

Docket No. 50-346

License No. NPF-3

Serial No. 1-494

January 18, 1985



RICHARD P. CROUSE  
Vice President  
Nuclear  
(419) 259-5221

Mr. James G. Keppler, Regional Administrator  
Region III  
United States Nuclear Regulatory Commission  
709 Roosevelt Road  
Glen Ellyn, Illinois 60137

Dear Mr. Keppler:

This letter transmits the information that Toledo Edison agreed to provide in the December 21, 1984 meeting at the Nuclear Regulatory Commission (NRC) Glen Ellyn offices, between members of the NRC Region III staff and Toledo Edison representatives, relative to a failed snubber found on the pressurizer surge line at the Davis-Besse Nuclear Power Station, Unit 1.

The listed information was requested prior to return to "full power" operation following the refueling outage.

- (a) Calculations to demonstrate postulated failure mechanism.

Calculation number M8, Revision C3, is attached, which models the failure mechanism for the failed snubber. The results of this calculation conclude that a transverse loading of 730 pounds with a deflection of .042 inches (assuming a stress concentration factor of 1.6) would cause failure. The postulated failure mechanism is supported by the calculation.

- (b) Calculations to demonstrate the adequacy of the Nuclear Class 1 stanchion to pipe weld.

The requested information is provided in the following attachments:

- (1) Calculation number C-PS/PSU-1;
- (2) Calculation number M8, Revision C2, in its entirety; and
- (3) Calculation number M8, Revision C1, Attachment 1, Sheet 1 of 1; Attachment 2, Sheet 2 of 4; Attachment 3, Sheets 2 of 3 and 3 of 3; and Attachment 4, Sheet 2 of 2.

These calculations show the stanchion to pipe weld to be acceptable for the design conditions.

- (c) Review of the pressurizer surge line, restraints and surrounding area to assure adequate clearances.

Field walkdowns and measurements were conducted on two occasions. The first on December 22, 1984 (Plant in Mode 6, Reactor Coolant System (RCS) temperature less than 150°F), and the second on January 11, 1985 (Plant in Mode 3, RCS temperature 535°F). The relative displacements of the pressurizer surge line at the failed snubber support location was: up (+7/8"); east (-1/4"); and north (-1 3/8").

The analyzed thermal movements shown on Attachment No. 7, Sheet 8 of 10, of Calculation number M8, Revision C2 were: up (+.177"); west (+.69"); and north (-1.651"). These movements consider the maximum operating conditions, hot leg at 608°F, and the pressurizer and surge line at 670°F.

Since the temperature of the RCS when the "hot" measurements were taken was less than the analyzed condition, the actual movements will be different. Bechtel performed a thermal analysis of the RCS, considering all the various operating and start-up conditions.

Results of this analysis show the surge line can move in the opposite direction and with magnitudes different from those originally considered. For example, when the hot leg is above 500°F, portions of the surge line can be at significantly lower temperatures due to minimal flow through the surge line and normal heat losses. This can result in line movement slightly east, as observed.

Accounting for the observed thermal movements and evaluating "worst case" combinations of start-up/operations, the range of movements does not invalidate the design of PSU-R1 as presently installed or lessen the interference with the wall that was postulated to cause failure of the snubber.

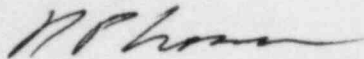
The field walkdowns also confirmed that adequate clearances are available for the pressurizer surge line, pressurizer spray line, and pressurizer sample line in the area in question.

Based on the above, Toledo Edison is confident of the adequacy of the new support design, which was completed during the refueling outage as corrective action relative to the failed snubber.

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Page 3

Toledo Edison feels the information submitted in this letter fully addresses the concerns expressed by members of the NRC staff in the December 21, 1984 meeting, and, therefore, removes the constraint to proceed to full power operation.

Very truly yours,

A handwritten signature in cursive script, appearing to read "J. P. ...".

RPC:JKW:nlf  
cc: DB-1 NRC Resident Inspector



# CALCULATION COVER SHEET

Q

PROJECT Davis Besse Unit-1 JOB NO. 12501009 DISCIPLINE MECH./PROCESS

SUBJECT Pipe Stress Analysis FILE NO. \_\_\_\_\_

ORIGINATOR William Gallagher CALC NO. m 8

CHECKER AL. Noy DATE 7/15/82

NO OF SHEETS 60

CALC. SHTS. = 23, ATTACHMENT # 1 = 12 WGS, ATTACHMENT # 2 = 4 SHTS

RECORD OF ISSUES ATTACHMENT # 3 = 3 SHTS, ATTACHMENT # 4 = 2 SHTS, ATTACHMENT # 5 = 3 SHTS, ATTACHMENT # 6 = 2 SHTS, ATTACHMENT # 7 = 10 SHTS

NO.	DESCRIPTION	BY	DATE	CHKD	DATE	APPRD	DATE	DATE FILMED
C1	ISSUED For Study	WFG	7/15/82	ALN	7/15/82	SPL	7-15-82	
C2	REVISED SHT NO. 3 AND ADDED SHT. NOS. 13, 14, 15, 16 & 17 AND ATTACH. NO. 617	SPL	12-5-84	SPO	12-5-84		12-5-84	
C3	ADDED SHT. NOS. 18 THRU 23 & ATTACHMENTS 8 THRU 13	WFG	12-27-84	WHL	12-27-84	MSW	12-28-84	

PRELIMINARY CALC.  COMMITTED PRELIMINARY DESIGN CALC.   
 SUPERSEDED CALC.  FINAL CALC.

Attachment # 8 = 3 SHTS Attachment # 9, 10, 11 & 12 = 1 SHT Attachment # 13 = 4 SHTS

## STUDY For FCR 82-013

ONE OF THE VERTICAL SNUBBERS FOR HANGER NO. PSU-R1 (DATA PT. 40) FAILED (NCR NO. 84-190).

IN ADDITION, THERE IS INTERFERENCE PROBLEM BETWEEN PIPE AND WALL AT THIS HANGER DUE TO THERMAL EXPANSION OF PIPE.

THIS STUDY IS MADE WITH THESE CONDITIONS.  
 NOTE: 1) REV. C2 IS INDEPENDENT OF REV. C1  
 2) REV. C2 → NCR NO. 84-190.

THIS REVISION MODELS THE FAILURE MECHANISM FOR THE BROKEN SNUBBER IN NCR 84-190

NOTE: REV. C3 IS INDEPENDENT OF REV'S C1 & C2  
 Rev. C3 adds shts. 18 thru 23 and Attachments 8 thru 13.





# CALCULATION SHEET

JOB NO. 12501	CALC. NO. M-8	REV. NO. C3	SHEET NO. 18
ORIGINATOR C. H. Abertin	DATE 12/27/84	CHECKED [Signature]	DATE 12-27-84

THIS CALCULATION IS TO DETERMINE THE MAX APPLIED LOAD THAT CAUSED THE SNUBBER TO BREAK AT THE CRITICAL POINT (PISTON ROD) & THE DEFLECTION NECESSARY TO INDUCE THIS LOAD

DATA FROM STANUL OUTPUT DATED 12/27/84

CHECK MEM 4 (MODELED FOR PISTON ROD - FAILURE AT JOINT 4)

$$F_b = \frac{M}{S} = \frac{2.625 \text{ kip}}{.0202 \text{ in}^3} = 111.386 \text{ ksi}$$

PISTON ROD REF: GRUNDY LCD = MATERIAL ASTM A 434-64  
1 1/2" FIG. 200 GR BC

ULTIMATE TENSILE STRENGTH = 130 ksi (REF. ATTACHMENT 13)

$$\text{APPLIED LOAD} = \frac{130 \text{ ksi}}{111.386 \text{ ksi}} (1.0 \text{ kip}) = 1.167 \text{ kips}$$

$$\text{FAILURE LOAD} = \frac{1.167 \#}{\text{STRESS CONC. FACTOR } K} = \frac{1.167 \#}{1.6} = 730 \#$$

(ref sheet 21)

$$\text{MAX. DEF. AT POINT OF LOAD} = \frac{730 \# (.0579)}{1000 \#}$$

$$\Delta_2 = .042''$$



# CALCULATION SHEET

JOB NO. 12E01	CALC. NO. M-8	REV NO. C3	SHEET NO. 19
ORIGINATOR C. H. Schubert	DATE 12/26/84	CHECKED J. E. U's	DATE 12-27-84

REF: FORMULAS FOR STRESS & STRAIN BY ROARK

$$r = \frac{.625}{2} = .3125"$$

$$\alpha = 36.87^\circ$$

$$\text{Arc Length} = \frac{\pi r 73.74^\circ}{180^\circ} = .402"$$

$$c = 2\sqrt{2br - b^2}$$

$$= 2\sqrt{2(.0625)(.3125) - (.0625)^2}$$

$$= .375 \text{ in}$$

$$A_x = A_y = A_z = \pi r^2 - \frac{1}{2}[(\text{ARC LENGTH} \times r) - c(r - b)]$$

$$= \pi(.3125)^2 - [(.402 \times .3125) - .375(.3125 - .0625)]$$

$$= .307 - [.126 - .09375]$$

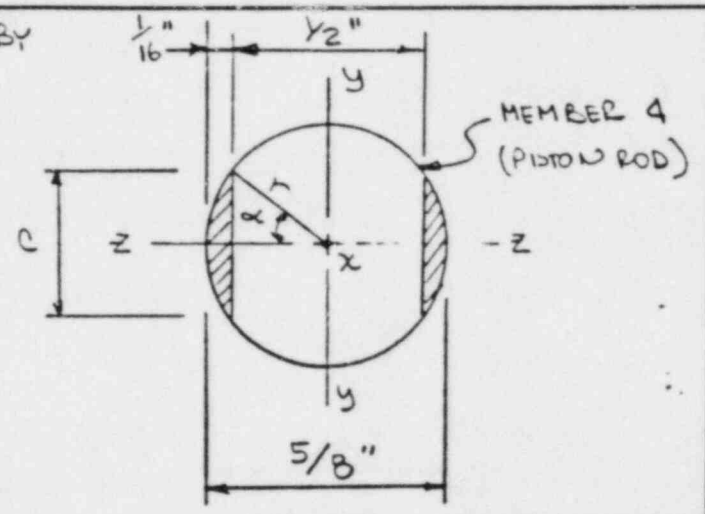
$$= .275 \text{ in}^2$$

$$I_{\text{circle}} = \frac{\pi (.3125)^4}{4} = .0075 \text{ in}^4$$

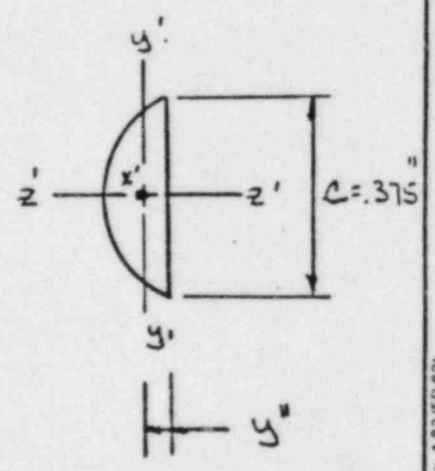
$$A' = \frac{.126 - .09375}{2} = .0161 \text{ in}^2$$

$$y'' = r \left[ \frac{2 \sin^3 \alpha}{3(\alpha - \sin \alpha \cos \alpha)} - \cos \alpha \right]$$

$$\alpha = \frac{36.87^\circ}{57.28^\circ/\text{RAD}} = .6437 \text{ RAD}$$



NOTE: REF: FORMULAS FOR STRAIN & STRESS TABLE I  
 $\alpha < \frac{\pi}{4}$  USE FORM OF SECTION NO. 18  
 FORM OF SECTION NO. 17 WAS USE IN CALC. NO SIGNIFICANT DIFFERENCE IN THIS CASE.  $\therefore$  OK.





# CALCULATION SHEET

JOB NO. 12501	CALC. NO. M-8	REV. NO. CB	SHEET NO. 20
ORIGINATOR C. H. Aburton	DATE 12/26/84	CHECKED [Signature]	DATE 12.27.84

$$y'' = .3125 \left[ \frac{2 \sin^3 36.87}{3(.6437 - \sin 36.87 \cos 36.87)} - \cos 36.87 \right]$$

$$= .3125 \left( \frac{.432}{.491} - .80 \right)$$

$$= .025''$$

$$Iy' - y' = \frac{r^4}{4} \left[ \alpha - \sin \alpha \cos \alpha + 2 \sin^3 \alpha \cos \alpha - \frac{16 \sin^6 \alpha}{9(\alpha - \sin \alpha \cos \alpha)} \right]$$

$$= \frac{(.3125)^4}{4} [ .6437 - .48 + .3456 - .507 ]$$

$$= .0024 (.0025)$$

$$= 5.5 \times 10^{-6} \text{ in}^4$$

$$Iy - y = .0075 - 2 [ .0000055 + .0161 (.025 + .25)^2 ]$$

$$= .00505 \text{ in}^4$$

$$S_y - y = \frac{.00505}{.25} = .0202 \text{ in}^3$$

$$Iz' - z' = \frac{r^4}{12} [ 3\alpha - 3 \sin \alpha \cos \alpha - 2 \sin^3 \alpha \cos \alpha ]$$

$$= .0008 [ 3 \times .6437 - 1.44 - .3456 ]$$

$$= .00012 \text{ in}^4$$

$$Iz - z = .0075 - 2(.00012)$$

$$= .00726 \text{ in}^4$$

$$S_z - z = \frac{.00726}{.3125} = .0232 \text{ in}^3$$

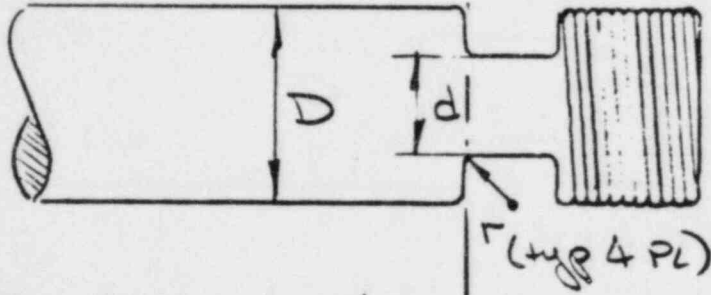


# CALCULATION SHEET

JOB NO. 12501	CALC. NO. U-8	REV. NO. 03	SHEET NO. 21
ORIGINATOR E. Cies	DATE 12-27-84	CHECKED C.H. [unclear]	DATE 12/27/84

PISTON SHAFT - ITT GERRINELL  
 FRACTURE PLANE ↓

1/2" FIG. 800  
 HYDRAULIC SNUBBER



STRESS CONCENTRATION  
 FACTOR (K)

$D = 5/8"$   $d = 1/2"$

1) ratio  $D/d = .625/.5 = 1.25$

approximate  $r = 1/16"$

$r/d = \frac{.0625}{.5} = .125 \rightarrow K = 1.6$

2) approximate  $r = 1/32"$

$r/d = \frac{.03125}{.5} = .0625 \rightarrow K = 2.1$

(ATTACHMENT 9)  
 FIG. 4.27  
 pg. 173

Mechanics of  
 Materials -  
 Beer, Johnston

1)  $D/d = 1.25$   $r/d = .125 \rightarrow K = 1.62$

2)  $r/d = .0625 \rightarrow K = 1.98$

(ATTACHMENT 10)  
 FIG. AF 9  
 pg. 585

1)  $D/d = 1.25$   $r/d = .125 \rightarrow K = 1.58$

2)  $r/d = .0625 \rightarrow K = 1.88$

(ATTACHMENT 11)  
 FIG. AF 12  
 pg. 586  
 Design of Machine  
 Elements -  
 Failes





# CALCULATION SHEET

JCB NO. 17501	CALC NO. N-8	REV. NO. C3	SHEET NO. 22
ORIGINATOR Kis	DATE 12 27 84	CHECKED O H Johnson	DATE 12/27/84

As can be seen, if we model the round shaft w/a flat slot as a flat bar (Attachments 9 - 10) or as a round shaft (Attachment 11) the stress concentration factor only varies from  $K=1.6 \rightarrow 2.1$

For purposes of this analysis, a lower  $K$  will give a higher load and deflection to cause failure, therefore use  $K=1.6$ .

## Conclusion -

From revision C2 it was shown that a thermal interference of .292" would have generated a reaction of 5288#. This revision shows that it would have only taken a reaction of 730# and a thermal interference of .042" to cause failure of the piston rod. From the scale drawing in Attachment 12 and from photographs of the wall it can be seen



# CALCULATION SHEET

JOB NO. 12501	CALC. NO. M-8	REV. NO. 03	SHEET NO. 23
ORIGINATOR 1 EUS	DATE 12-27-84	CHECKED C.H. Akers	DATE 12/27/84

1  
2 That at the upper tube steel contacted the 45°  
3 haunch of the wall after approximately 3/8"  
4 of movement. The 45° incline then forced the  
5 Surge line down during continued west thermal  
6 movement. When the tube steel finally contacted the  
7 vertical portion of the wall just below the haunch there  
8 had already been an interference between the two (?)  
9 valve body bolts and the wall. The effect of this  
10 interference can be seen on the photographs as  
11 chipped concrete in a circular arc ≈ 1" long. The  
12 magnitude of the interference can be approximated as  
13 ≈ 1" / 1.6" × .667" ≈ .42" where ≈ 1" is the length of contact with  
14 wall in a north direction, 1.6" is the total analyzed north mit,  
15 and .667" is total analyzed west mit. This interference  
16 with a relative east-west piping system stiffness of  
17 5288 / .292 = 18109 #/in would generate a force of 7506 #,  
18 obviously much greater than the necessary 730 #  
19  
20  
21  
22  
23  
24  
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34  
35  
36

```

*****
*
*   UNIVAC 1100 SERIES ICES
*   DEC 27, 1984           07:57:12
*   VERSION 2.9
*
*****

```

>@ADD CAR2.

1. STRUDL 'PSU-R1'

```

*****
*
*   ICES STRUDL-II
*   THE STRUCTURAL DESIGN LANGUAGE
*
-----
*   CIVIL ENGINEERING SYSTEMS LABORATORY
*   MASSACHUSETTS INSTITUTE OF TECHNOLOGY
*   CAMBRIDGE, MASSACHUSETTS
*
*   07.57.46           27 DEC 84
*   SPERRY UNIVAC 1100 SERIES
*   UNIVAC 1100 SERIES EXEC 8
*   VERSION 2.9.2
*
*****

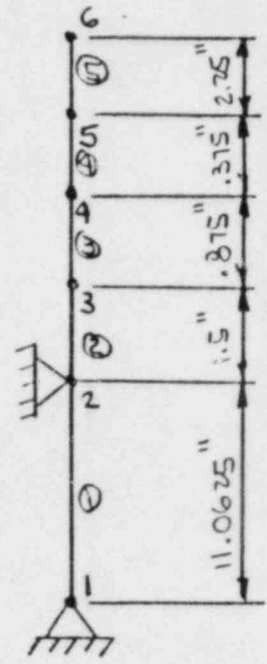
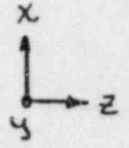
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2. TYPE SPACE FRAME
3. UNIT IN KIP DEG
4. CONSTANT E 27100. ALL
5. BETA 0. ALL
6. JOINT COORDINATES
7. 1      0      0      0      SUPPORT
8. 2      11.0625 0      0      SUPPORT
9. 3      12.5625 0      0
10. 4      13.4375 0      0
11. 5      13.8125 0      0
12. 6      16.0625 0      0
13. MEMBER INCIDENCES
14. 1      1      2
15. 2      2      3
16. 3      3      4
17. 4      4      5
18. 5      5      6
19. JOINT RELEASE
20. 1 MOMENT Y
21. 2 MOMENT Y
22. MEM PROP
23. 1 2 5 PRIS AX 10 AY 10 AZ 10 IX 10 IY 10 IZ 10
24. 3 PRIS AX .307 AY .307 AZ .307 IX .015 IY .0075 -
25. IZ .0075 SY .024 SZ .024
26. 4 DDIS AY 275 AZ 275 AX 275 AY 275 IY 01015 IY 00505 -

```

ORIGINATOR: O. H. Rubin CHECKED BY: HT US DATE: 12 27 84



JOB #12501  
ATTACHMENT # 8, slt 1/3  
CALC. # M-8  
REV. C3

- 28. LOADING 1
- 29. JOINT LOAD
- 30. & FORCE Z -1.0
- 31. STIFFN A
- 32. OUTPUT BY MEMBER
- 33. OUT DEC 4
- 34. LIST FORCE REA DISP ALL

SHOULD BE :  
 \* I<sub>z</sub> = .00726 } OUT PUT NOT  
 S<sub>z</sub> = .0232 } AFFECTED CMA 12/21/24

\*\*\*\*\*  
 \*RESULTS OF LATEST ANALYSIS\*  
 \*\*\*\*\*

PROBLEM - PSU-R1 TITLE -

ACTIVE UNITS INCH KIPS DEGREE DEGF SECOND

ACTIVE STRUCTURE TYPE SPACE FRAME

ACTIVE COORDINATE AXES X Y Z

JOB #12501  
 ATTACHMENT # 8-2/2/3  
 CALC. # M-8  
 REV. C3

MEMBER FORCES

MEMBER	LOADING	JOINT	FORCES			MOMENTS		
			AXIAL	SHEAR Y	SHEAR Z	TORSIONAL	BENDING Y	BENDING Z
1	1	1	.0000	.0000	-.4520	.0000	.0000	.0000
		2	.0000	.0000	.4520	.0000	5.0000	.0000
2	1	2	.0000	.0000	1.0000	.0000	-5.0000	.0000
		3	.0000	.0000	-1.0000	.0000	3.5000	.0000
3	1	3	.0000	.0000	1.0000	.0000	-3.5000	.0000
		4	.0000	.0000	-1.0000	.0000	2.6250	.0000
4	1	4	.0000	.0000	1.0000	.0000	-2.6250	.0000
		5	.0000	.0000	-1.0000	.0000	2.2500	.0000
5	1	5	.0000	.0000	1.0000	.0000	-2.2500	.0000
		6	.0000	.0000	-1.0000	.0000	.0000	.0000

JOINT SUPPORTS



JOINT		FORCES			MOMENTS		
LOADING	X FORCE	Y FORCE	Z FORCE	X MOMENT	Y MOMENT	Z MOMENT	
1	.0000	.0000	-.4520	.0000	.0000	.0000	
2	.0000	.0000	1.4520	.0000	-.0000	.0000	

JOINT SUPPORTS

JOINT		DISPLACEMENTS			ROTATIONS		
LOADING	X DISPL	Y DISPL	Z DISPL	X ROT	Y ROT	Z ROT	
1	.0000	.0000	.0000	.0000	-.0017	.0000	
2	.0000	.0000	.0000	.0000	.0041	.0000	

JOINT FREE JOINTS

JOINT		DISPLACEMENTS			ROTATIONS		
LOADING	X DISPL	Y DISPL	Z DISPL	X ROT	Y ROT	Z ROT	
3	.0000	.0000	-.0001	.0000	.0055	.0000	
4	.0000	.0000	-.0065	.0000	.7609	.0000	
5	.0000	.0000	-.0129	.0000	1.1436	.0000	
6	.0000	.0000	→ -.0579	.0000	1.1441	.0000	

35. FINISH

CORE REQUESTED 62074 DECIMAL WORDS  
THERE WERE 6 REORGANIZATIONS IN THIS JOB

- 1 DATA COMPACTIIONS
- 1 MODULE COMPACTIIONS
- 0 LOW RELEASED
- 4 LOW DEAD MODULES
- 0 HIGH RELEASED
- 0 HIGH DEAD MODULES
- 0 LOW UNRELEASED
- 24 DD4 BLOCKS READ FROM BANKS

JOB # 12501
ATTACHMENT # 8 5/13/3
CALC. # N-8
REV. CB

4.7. STRESS CONCENTRATIONS

The formula  $\sigma_m = Mc/I$  was derived in Sec. 4.4 for a member with a plane of symmetry and a uniform cross section, and we saw in Sec. 4.5 that it was accurate throughout the entire length of the member only if the couples  $M$  and  $M'$  were applied through the use of rigid and smooth plates. Under other conditions of application of the loads, stress concentrations will exist near the points where the loads are applied.

Higher stresses will also occur if the cross section of the member undergoes a sudden change. Two particular cases of interest have been studied,† the case of a flat bar with a sudden change in width, and the case of a flat bar with grooves. Since the distribution of stresses in the critical cross sections depends only upon the geometry of the members, stress-concentration factors may be determined for various ratios of the parameters involved and recorded as shown in Figs. 4.27 and 4.28. The value of the

ATTACHMENT 9  
 CALC N-8 RECS  
 By K. P. L. S. 12-27-84  
 Chk C. H. Oberlin 12/27/84

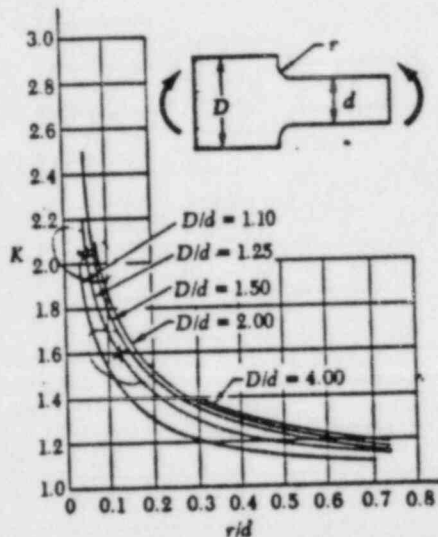


Fig. 4.27 Stress-concentration factors for flat bars with fillets under pure bending.†

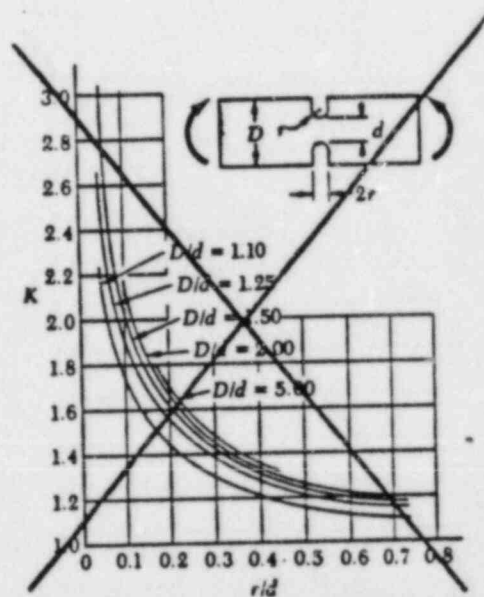


Fig. 4.28 Stress-concentration factors for flat bars with grooves under pure bending.†

† M. M. Frocht, "Photoelastic Studies in Stress Concentration," *Mechanical Engineering*, August 1936, pp. 485-489.

ATTACHMENT 10  
 CALC N-S 60 C3  
 By        12-27-84  
 Cld        12/27/84

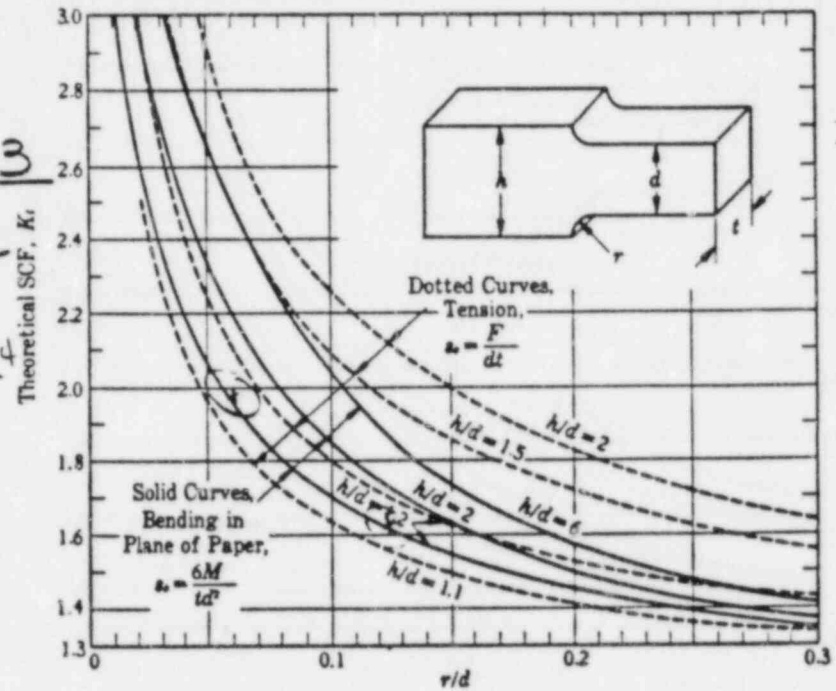


FIGURE AF 9 Flat Plate with Fillets. The tensile load is central. For  $h/d = 1.1$ , tensile and bending curves are very close together down to  $r/d = 0.04$ . (After R. E. Peterson)<sup>(4, 21)</sup>

particularly to normal  
 4.57)

and Bending (4.5, 4.16, 4.21)  
 ) with a hole ( $d$ ) in tension.  
 ile rectangular body are

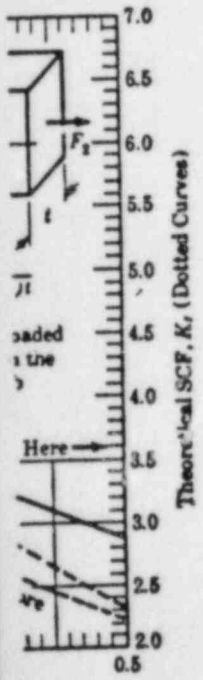
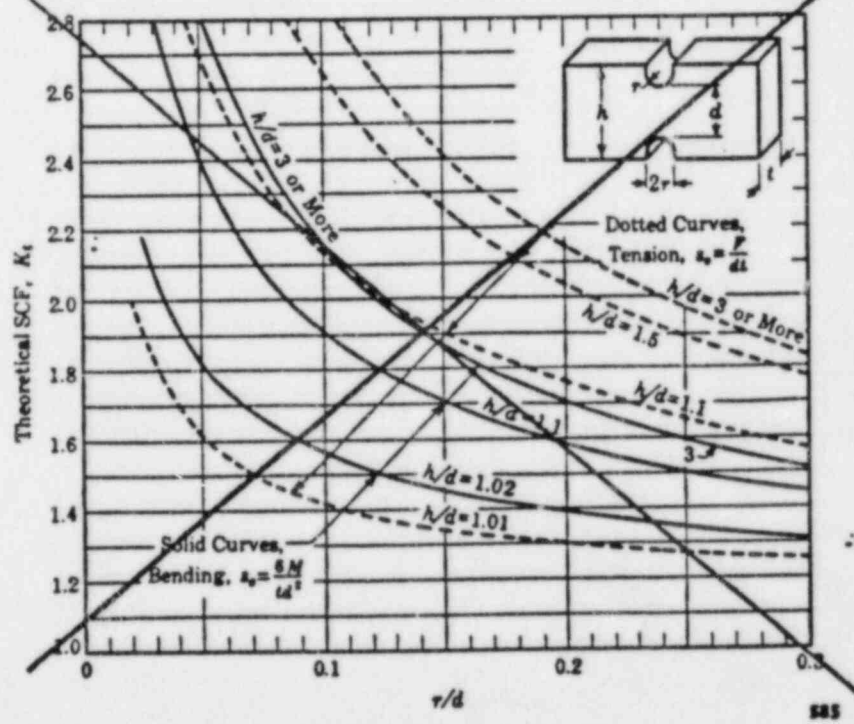


FIGURE AF 10 Flat Plate with Grooves.





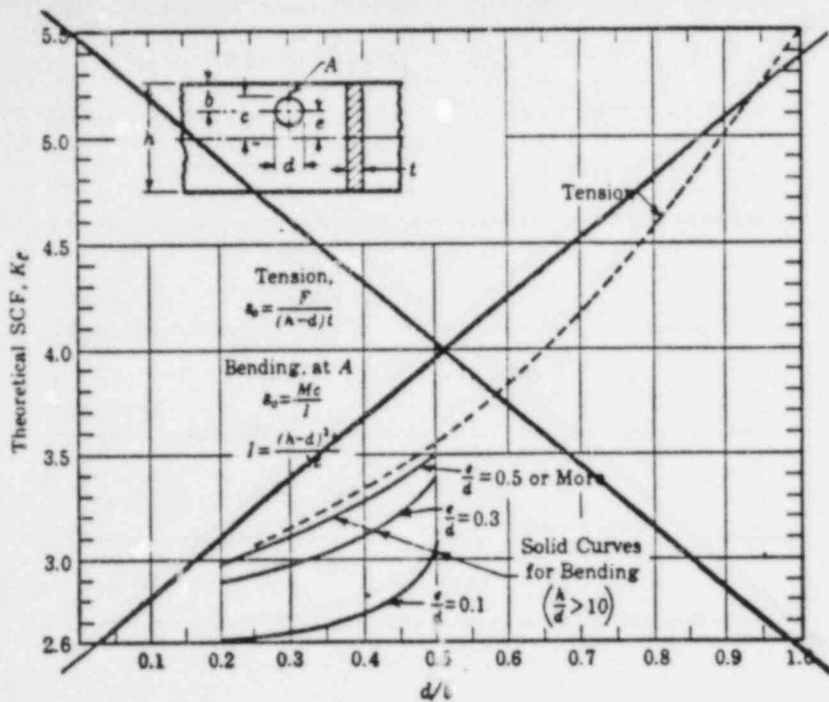
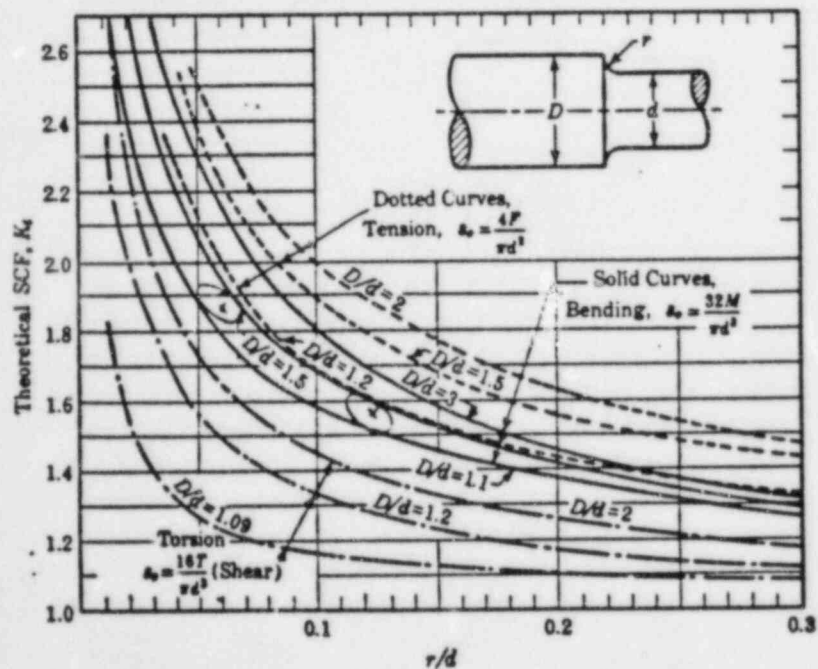


FIGURE AF 11 Flat Plate with Eccentric Hole. (4.2.4.20) For tension, if  $h < 20d$ ,  $K_t$  is somewhat smaller than shown. (After R. E. Peterson).<sup>(4.21)</sup>

FIGURE AF 12 Shaft with Fillet. The tensile load is central. Torsion curve  $D/d = 1.2$  approximates the bending curve for  $D/d = 1.01$ ; torsion curve  $D/d = 2$ , approximates the bending curve for  $D/d = 1.02$  (down to  $r/d \approx 0.04$ ). Bending curve  $D/d = 1.1$  approximates the tension curve  $D/d = 1.1$ . (After R. E. Peterson).<sup>(4.21)</sup>



ATTACHMENT 11  
 CALC M-8 60C3  
 By 12/27/54  
 Cled C.H. Chubb 12/27/54

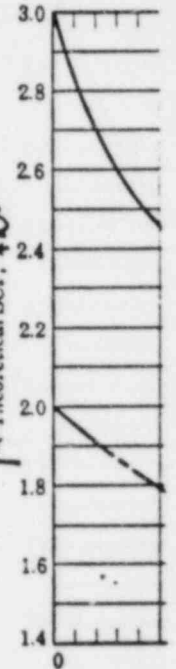
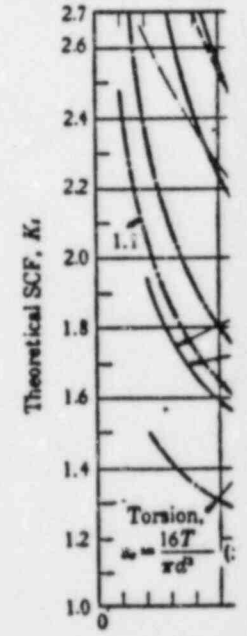


FIGURE AF 13 Shaft with Radial stress for this case falls slightly inside some point B. (After R. E. Peterson)

FIGURE AF 14 Shaft with Groove, as bending (approximate).





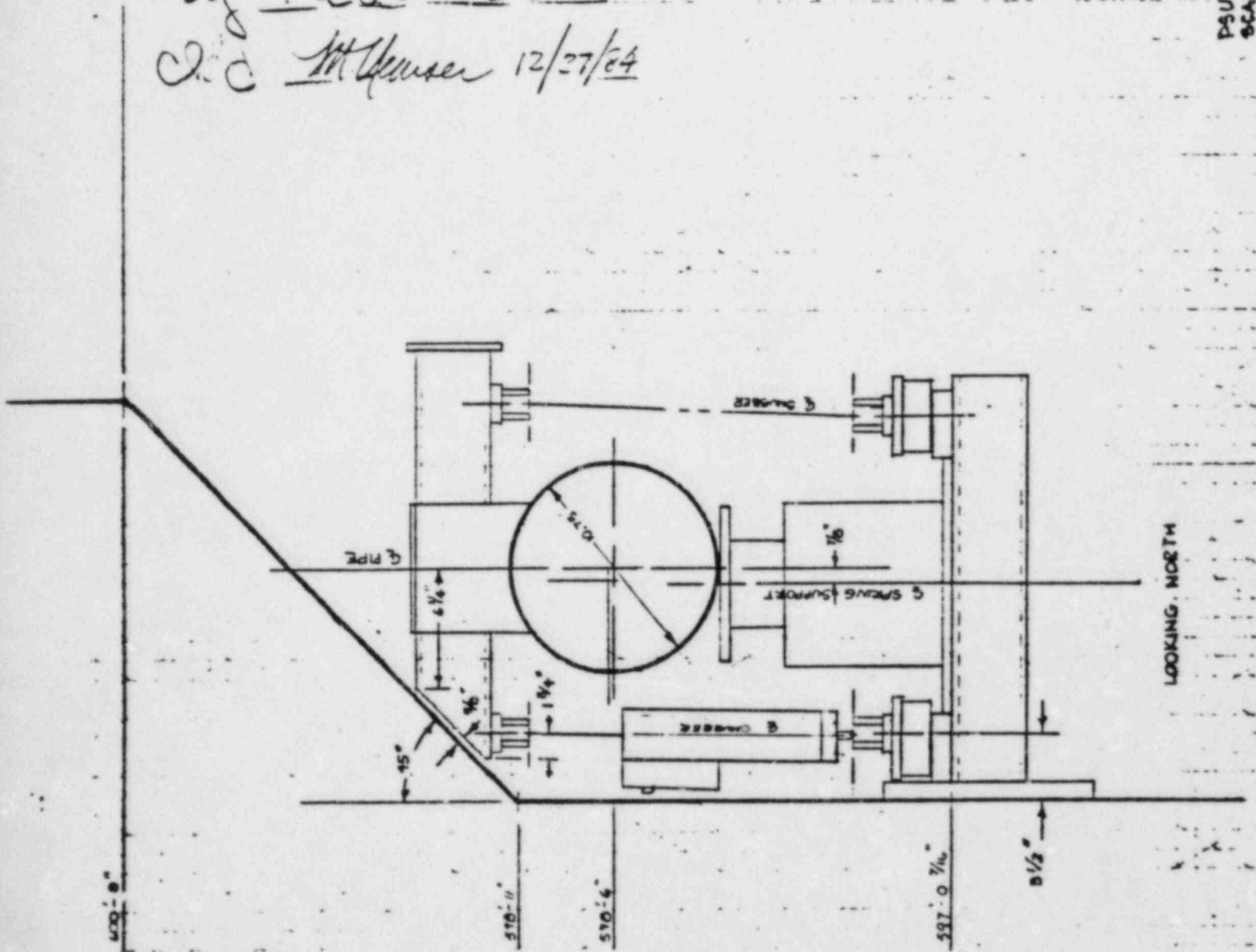
ATTACHMENT 12

CALC. M-8 REV. C3

By DEW 12-27-84

Ch. C M. Hansen 12/27/84

PSU-R1  
SCALE: 7/8" = 1"



ATTACHMENT 13 SHT 1 of 4  
 Calc M-8 rev. C3

PAGE 2 OF 6  
 FIG. 200, 201  
 LCDS REV. 7

By Dies 12/27/84 ITT GRINNELL  
 Dec 12/27/84 PIPE HANGER DIVISION  
 ASME SECTION III  
 SUBSECTION NF  
 LOAD CAPACITY DATA SHEET

4. COMPONENT SUPPORT INFORMATION		
ITEM	MATERIAL SPECIFICATION	ANALYSIS
REAR BRACKET BASE PLATE	SA-515 GR 65 or SA-36	PLATE & SHELL
REAR BRACKET LUGS	SA-515 GR 65 or SA-36	LINEAR
REAR BRACKET PIN	SA-193 GR B 7	LINEAR
PIVOT LUG	SA-515 GR 65 & SA-36	LINEAR
PIVOT LUG BASE PLATE	SA-515 GR 65 & SA-36	PLATE & SHELL
CYLINDER CAP	ASTM A 108-73 GR 1018*	PLATE & SHELL
CYLINDER BARREL	ASTM A 519-74 GR 1018 CW*	PLATE & SHELL
TIE RODS	ASTM A-193-73 GR B 7	LINEAR
TIE ROD NUTS	ASTM A 194-73 GR 7	LINEAR
PISTON (1 1/2 - 6")	ASTM A 108-73 GR 1018*	LINEAR
PISTON (8")	ASTM A434-64 CLASS BC	LINEAR
PISTON ROD	ASTM A 434-64 GR B C*	LINEAR
CYLINDER ROD EYE	SA-515 GR 65, SA-36, SA-182 GR F22	LINEAR
PIPE CLAMP (DESIGN TEMP = 650 F)	SA-515 GR 65, SA-36	LINEAR
CLAMPING BOLTS (DES. TEMP = 650 F)	SA-307 GR B*, or ASTM A-307-74 GR A*	LINEAR
CLAMP LOAD STUD	SA-193 GR B 7	LINEAR
EXTENSION ROD END (FIG. 201 ONLY)	SA-515 GR 65, SA-36, SA-675 GR 50-70	LINEAR

TABLE 1 Tensile Properties

NOTE—The mechanical properties of hot-wrought bars ordered for cold finishing shall be governed by the cold-finished size

Class	Diameter, in. (mm)	Yield Strength, min. ksi (MPa)*	Tensile Strength, min. ksi (MPa)	Elongation in 2 in. or 50 mm, min. %	Reduction of Area, min. %	SAE or AISI Representative Grade Types*
BB <sup>c</sup>	1½ (38.1) and under	90 (620)	110 (760)	20	50	3100, 4100, 8600, 8700
	Over 1½ to 2½ (38.1 to 63.5), incl	80 (550)	105 (720)	20	50	
	Over 2½ to 4 (63.5 to 114), incl	75 (520)	100 (690)	20	50	
	Over 4 to 7 (114 to 178), incl	75 (520)	95 (660)	20	50	
	Over 7 to 9½ (178 to 241.3), incl	65 (450)	90 (620)	18	40	
BC <sup>c</sup>	1½ (38.1) and under	110 (760)	130 (900)	16	50	3100, 4100, 6100, 8600, 8700
	Over 1½ to 2½ (38.1 to 63.5), incl	105 (720)	125 (860)	16	50	
	Over 2½ to 4 (63.5 to 114), incl	95 (660)	115 (790)	16	45	
	Over 4 to 7 (114 to 178), incl	85 (590)	110 (760)	16	45	
	Over 7 to 9½ (178 to 241.3), incl	80 (550)	105 (720)	15	40	
BD <sup>c</sup>	1½ (38.1) and under	130 (900)	155 (1070)	14	35	4100, 4300, 9800
	Over 1½ to 2½ (38.1 to 63.5), incl	120 (830)	150 (1030)	14	35	
	Over 2½ to 4 (63.5 to 114), incl	110 (760)	140 (960)	14	35	
	Over 4 to 7 (114 to 178), incl	105 (720)	135 (930)	14	35	
	Over 7 to 9½ (178 to 241.3), incl	100 (690)	130 (900)	14	35	

\* Determined by the 0.2% offset method.  
 \* The carbon content of all steels listed in this column may vary up to 0.55% max. as agreed upon by the manufacturer and the purchaser.  
<sup>c</sup> Class BB, BC, BD hot-wrought; Class BB, BC cold-finished.

The American Society for Testing and Materials takes no position respecting the validity of any patent rights asserted in connection with any item mentioned in this standard. Users of this standard are expressly advised that determination of the validity of any such patent rights, and the risk of infringement of such rights, are entirely their own responsibility.

This standard is subject to revision at any time by the responsible technical committee and must be reviewed every five years and if not revised, either reapproved or withdrawn. Your comments are invited either for revision of this standard or for additional standards and should be addressed to ASTM Headquarters. Your comments will receive careful consideration at a meeting of the responsible technical committee, which you may attend. If you feel that your comments have not received a fair hearing you should make your views known to the ASTM Committee on Standards, 1916 Race St., Philadelphia, Pa. 19103.

ATTACHMENT 13 Sheet 2 of 4  
 CALC M-8 rev. C3  
 By 1/2/84 12-27-84  
 J.C.D. M. G... 12/27/84

Standard  
MAGNE  
FORGIN

This standard is issued under original adoption or, in the A superscript epsilon (ε) in.

1. Scope

1.1 This is an acc magnetic particle i crankshafts having crankpins 8 in. (203

1.2 There are thr standards of increasi  
 1.2.1 Class 1  
 1.2.2 Class 2 (ori standard of Specifica  
 1.2.3 Class 3 (forr tary requirement S1 (1970)).

1.3 This specifica diesel locomotive c Class 2 or 3 may be ultimate user agree t  
 NOTE—Specification bon and Alloy, for Ger specification which ma

1.4 The values st to be regarded as th

2. Applicable Docu  
 2.1 ASTM Stand A 275 Method fc nation of Steel E 165 Recommen erant Inspectio

3. Ordering Inform.  
 3.1 The inquiry shall contain a statu s (are) to be subjec this specification. If Class 2 shall apply.

TABLE 3 (Cont'd)  
Yield Strength Values,  $S_y$ , for Ferritic Steels and Copper Alloys for Classes 1, 2, 3, and MC Linear Type Component Supports

Nominal Composition	P- No.	Group No.	Product Form	Specifica- tion No.	Type or Grade	Class	Notes	Min Yield Strength	Min Ultimate Tensile Strength	Yield Strength, ksi (multiply by 1000 to obtain psi) for metal temperatures, F, not to exceed									
										100	200	300	400	500	600	650	700	750	800
AISI 4140, 4145, 4340			Bar	A434-64	-	BC	1,2	110	130	110.0	102.9	99.4	96.6	93.6	89.5	86.8	83.6	-	-
			Bar	A434-64	-	BC	1,2	105	125	105.0	98.1	94.8	92.2	89.4	85.5	82.9	79.8	-	-
			Bar	A434-64	-	BC	1,2	95	115	95.0	88.5	85.4	83.0	80.6	77.0	74.6	72.0	-	-
			Bar	A434-64	-	BC	2	85	110	85.0	79.5	76.8	74.6	72.4	69.1	67.0	64.6	-	-
			Bar	A434-64	-	BC	2	80	105	80.0	74.8	72.3	70.3	68.1	65.1	63.1	60.9	-	-
AISI 4140, 4145, 4340			Bar	A434-64	-	BD	1,2	130	155	130.0	121.5	117.2	114.1	110.7	105.7	102.5	98.8	-	-
			Bar	A434-64	-	BD	1,2	120	150	120.0	112.1	108.4	105.2	102.1	97.6	94.6	91.1	-	-
			Bar	A434-64	-	BD	1,2	110	140	110.0	102.9	99.4	96.6	93.6	89.5	86.8	83.6	-	-
			Bar	A434-64	-	BD	1,2	105	135	105.0	98.1	94.8	92.2	89.4	88.5	82.9	79.8	-	-
			Bar	A434-64	-	BD	1,2	100	130	100.0	93.5	90.3	87.8	85.1	81.4	78.9	76.0	-	-
			Bar	SA-540	B21,B22, B23,B24	1	1,2	150	165	150.0	140.1	135.3	131.7	127.7	122.6	118.2	114.0	-	-
			Bar	SA-540	B21,B22, B23,B24	2	1,2	140	155	140.0	131.0	126.3	123.0	119.1	114.0	110.4	106.3	-	-
			Bar	SA-540	B21,B22, B23,B24	3	1,2	130	145	130.0	121.5	117.2	114.1	110.7	105.7	102.5	98.8	-	-
			Bar	SA-540	B21,B22, B23,B24	4	1,2	120	135	120.0	112.1	108.4	103.2	102.1	97.6	94.6	91.1	-	-
			Bar	SA-540	B21,B22, B23,B24	5	1,2	100	115	100.0	93.5	90.3	87.8	85.1	81.4	78.9	76.0	-	-
			Bar	SA-540	B21,B22, B23,B24	5	1,2	105	120	105.0	98.1	94.8	92.2	89.4	85.5	82.9	79.8	-	-
5Ni-Cr-Mo-V			Plate, Bar	A572-74b	-	-	-	42	60	42.0	40.0	38.3	36.8	35.2	33.5	32.7	31.8	-	-
			Plate, Bar	A572-74b	-	-	-	50	65	50.0	47.5	45.6	43.8	41.8	39.9	38.9	37.9	-	-
			Forging	A579-70	12c*	-	1,2	140	150	140.0	138.6	134.0	129.5	127.7	126.3	123.5	117.6	-	-
			Plate, Bar	A588-74a	A,B	-	-	42	63	42.0	40.0	38.3	36.8	35.2	33.5	32.7	31.8	-	-
			Plate, Bar	A588-74a	A,B	-	-	46	67	46.0	43.8	41.9	40.3	38.6	36.7	35.8	34.8	-	-
			Plate, Bar	A588-74a	A,B	-	-	50	70	50.0	47.5	45.6	43.0	41.8	39.9	38.9	37.9	-	-
			Tube	A618-74	II	-	-	50	70	50.0	-	-	-	-	-	-	-	-	-
			Tube	A618-74	III	-	-	50	65	50.0	-	-	-	-	-	-	-	-	-

\*Grade 12 modified

A-1000-13 SUR 3 of 4  
 CASE M-8, REV C3  
 SURFACES - 12-27-84  
 12/27/84  
 CASES OF ASME BOILER AND PRESSURE VESSEL CODE

CASE (continued)  
 1644-6



ATTACHMENT 13  
 SHEET 4 of 4  
 CALC M-8, REV. C3  
 By [Signature] 12-27-84  
 Ch E [Signature] 12/27/84

APPENDIX I

Table I-6.0

TABLE I-6.0  
 MODULI OF ELASTICITY OF MATERIALS FOR GIVEN TEMPERATURES

Material	Modulus of Elasticity, E, = Value Given × 10 <sup>6</sup> (psi) for Temperature (F) of										
	-325	-200	-100	70	200	300	400	500	600	700	800
<b>Ferrous Materials</b>											
Carbon steels with carbon content 0.30 or less, 3% Ni	30.0	29.5	29.0	27.9	27.7	27.4	27.0	26.4	25.7	24.8	...
Carbon steels with carbon content above 0.30	31.0	30.6	30.4	29.9	29.5	29.0	28.3	27.4	26.7	25.4	...
Carbon-molybdenum steels, low chrome steels through 3 Cr	31.0	30.6	30.4	29.9	29.5	29.0	28.6	28.0	27.4	26.6	...
Intermediate chrome steels (5 Cr through -9 Cr)	29.4	28.5	28.1	27.4	27.1	26.8	26.4	26.0	25.4	24.9	...
Austenitic steels (304, 310, 316, 321, 347)	30.4	29.9	29.4	28.3	27.7	27.1	26.6	26.1	25.4	24.8	24.1
Straight chromium steels (12 Cr, 17 Cr, 27 Cr)	30.8	30.3	29.8	29.2	28.7	28.3	27.7	27.0	26.0	24.8	23.1
<b>Nonferrous Materials</b>											
<b>High Nickel Alloys</b>											
Ni-Cr-Fe	...	...	...	31.7	30.9	30.5	30.0	29.6	29.2	28.6	27.9
Ni-Fe-Cr	...	...	...	...	...	...	...	...	...	...	...
Ni-Cu	...	...	...	29.0	28.4	27.9	27.5	27.1	26.7	26.3	25.8
Ni-Cr-Fe-Mo-Cb	...	...	...	...	...	...	...	...	...	...	...
<b>Other Nonferrous Materials</b>											
Copper Nickel (70-30)	...	...	...	22.0	...	...	...	...	...	...	...
Aluminum	11.3	11.1	10.9	10.6	10.3	10.2	9.5	8.5	...	...	...
Copper	17.0	16.7	16.5	16.0	15.6	15.4	15.1	14.7	14.2	13.7	...
Unalloyed Titanium	...	...	...	15.5	15.0	...	13.8	13.2	12.5	11.8	11.2

Assume temperature @ snubber = 550°F

$$E = \frac{27.4 + 26.7}{2} = 27.1 \times 10^6 \text{ psi}$$



# CALCULATION COVER SHEET

0

PROJECT DAVIS-BESSE UNIT 1 Discipline Civil/Pipe Support  
 SUBJECT DEPRESSURIZER SPRAY AND SURGE JOB NO. 12501 CALC. NO. C-PS/PSU-1

Iso. No. \_\_\_\_\_

COMPUTER PROGRAM:  NONE  SCP  OTHER  
 PROGRAM NO(S) see below VERSION/RELEASE NO. see below TOTAL NO. OF SHEETS 9

PRELIMINARY CALC.  COMMITTED PRELIMINARY CALC.  
 SUPERSEDES CALC. NO. \_\_\_\_\_  FINAL CALC.

STATEMENT OF PROBLEM  
 Design of Pipe Support for Structural Stability  
 Support No. PSU-H1, PSU-R1  
 For Calcs. See Calc. No. \_\_\_\_\_  
 No Calcs. Req'd. OK by Inspection

SAR CHECKED  SAR CHANGE REQ'D  SAR CHANGE NOTICE INITIATED

SOURCES OF DATA  
 Piping Stress Analysis Data - Prob. No. B 2 W DP. \_\_\_\_\_ Issue No. \_\_\_\_\_  
 ICES STRUDL II - (CE 901) - Version/Rel. No. \_\_\_\_\_ Output Attached \_\_\_\_\_  
 Anchor Bolt Prog. - (CE 035)-Version/Rel. No. \_\_\_\_\_ Output Attached \_\_\_\_\_  
 Anchor Bolt Prog. - (CE 050)-Version/Rel. No. \_\_\_\_\_  
 Local Stress Analysis - (ME 210) - Version/Rel. No. 5  
 STRESS GROUP MEMO ATTACHMENT A-1

SOURCES OF FORMULAE & REFERENCES  
 ASME III Subsection NF/Appendix XVII  
 ANSI B31.1  
 AISC Manual of Steel Construction - 7th Edition  
 Pipe Support Design Manual Vol. 1 & 2  
 Design of Welded Structures by O. W. Blodgett  
 GPD Pipe Support Group Newsletter No. \_\_\_\_\_

NO.	DESCRIPTION	BY	DATE	CHKD.	DATE	APPROVED	DATE
0	OPERABILITY FOR UCR BA-0180	CHA	12/9/84	T.H.W.	12/11/84	ME	12/17/84

### REVISIONS



# CALCULATION SHEET

ORIGINATOR Sura P. Rungano DATE 12.1.84 CALC. NO. C-PS/PSU- REV. NO. 0  
 PROJECT DAVIS-BESSE UNIT 1 CHECKED [Signature] DATE 12-5-84  
 SUBJECT DATA SHEET FOR LOCAL STRESS CHECK JOB NO. 125VT SHEET NO. 5

## FOR OPERABILITY

TO BE FILLED IN BY STRESS GROUP:

STRESS PROBLEM NO. M-8 ISSUE C2 DATA POINT 40  
 PIPE SUPPORT NO. PSU-R1

TYPE OF SUPPORT:

VERTICAL  (SNUBBER)  
 X  Z  ANCHOR

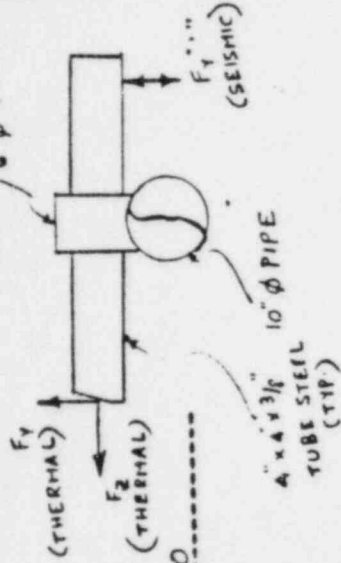
SKRU

ANCHOR

PIPE OUTSIDE DIAMETER 10.75 in  
 OPERATING PRESSURE 2750 psig  
 PIPING MATERIAL S.S. SA376-TP316

PIPE WALL THK 1.0 in  
 DESIGN PRESSURE 2500 psig  
 TEMPERATURE 670 °F

SM 16.48 ksi



ME-101 LOAD CASE	SUPPORT LOADS						PIPE STRESS (psi)
	FORCES (lbs)			MOMENTS (in-lbs)			
	X	Y	Z	X	Y	Z	
WEIGHT	-	2400					825
DBE		3900					4415
SSE		2615	5288				8203
THERMAL							4546
SAR							-

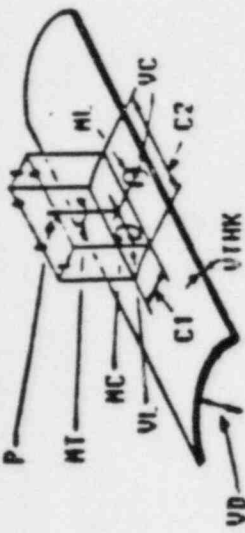
NOTE: 1) The stresses are at the center line of pipe.  
 2) The loads are as shown in figure.



# CALCULATION SHEET

ORIGINATOR C.H. Oberster DATE 12/4/84 CALC NO. C-PS/PSU-1 REV. NO. 0  
 PROJECT WALIS-BESSE UNIT 1 CHECKED T. LIN DATE 12/11/84  
 SUBJECT SE 210 INPUT DATA JOB NO. 12501 SHEET NO. 6

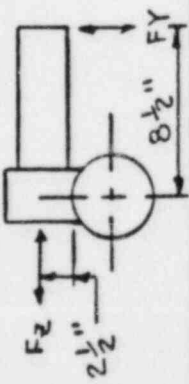
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-> TYPE CALL 816566\*PRG.210IM  
 1> TITLE, SUPPORT NUMBER PSU-R1  
 2> NUMBER OF LOADING CASES 3

### LOADING COMBINATIONS

CASE NUMBER	TYPE OF COMBINATION	P lbs	ML in-lbs	MC in-lbs	MT in-lbs	VL lbs	VC lbs	COMMENTS
1	WEIGHT+ DBE	2400 ;	0 ;	20400 ;	0 ;	0 ;	0	
2	WEIGHT+ SSE	3900 ;	0 ;	33150 ;	0 ;	0 ;	0	
3	WEIGHT+THRM+ ORE+SAM	5015 ;	0 ;	55848 ;	0 ;	0 ;	5288	



### GEOMETRY

PIPE WALL THICKNESS ACTUAL EFFECTIVE in	PIPE DIAMETER ACTUAL MEAN ADJUSTED in	ATTACHMENT DIMENSIONS		SHAPE
		C1 in	C2 in	
1.0 ;	13.25 ;	6.625 ;	6.625 ;	1.0
;	;	;	;	;

GEOMETRIC PARAMETERS  
 $\beta$  AND  $\gamma$  CHECK  
 $\beta = C1/VD = .5$   
 $\beta = C2/VD = .5$   
 $\gamma = \frac{VD}{12 \cdot VTHK} = 6.625$   
 $0.01 < \beta < 0.5$  IF NOT, ADJUST VD.  
 $5 < \gamma < 300$  ADJUST VD.  
 FOR POINT R:  
 $V^2 = \sqrt{VU^2 + VTU^2}$

7> TYPE +CALL 816566\*PRG.END  
 -> TYPE PXOT ADD 816566\*210RUM.

-> TYPE CALL 816566\*PRG.210SUB ADJUST PAPER, HIT RETURN



ORIGINATOR *O.H. Johnson*  
 PROJECT *DAVIS BESSE*  
*UNIT 1*

DATE *12/14/84*  
 JOB # *12501*

CHECKER *T. King* DATE *12/11/84*  
 STANDARD COMPUTER PROGRAM ME-210

CALC NUM *C-PS/PSIA-1*  
 REV NUM *0*

SHEET - 7 OF

VERSION : 5  
 RELEASED : DEC 15, 1982  
 USER MANUAL VERSION : 1  
 THEORETICAL MANUAL VERSION : 1  
 VERIFICATION MANUAL VERSION : 5

B I J L A A R D S T R E S S A N A L Y S I S F O R C Y L I N D E R S

PSU -H1,R1

I N P U T D A T A

	P (LBS)	ML (IN-LBS)	MC (IN-LBS)	MT (IN-LBS)	VL (LBS)	VC (LBS)	MA (FT-LBS)	MB (FT-LBS)
1	2400.	0.	20400.	0.	0.	0.	0.	0.
2	3900.	0.	33150.	0.	0.	0.	0.	0.
3	5015.	0.	55848.	0.	0.	5288.	0.	0.

	VESTHK (IN)	VESDIA (IN)	C1 (IN)	C2 (IN)	SHAPE	PRESSURE (PSI)	SM (KSI)	BEND R (IN)
1	1.0000	13.2500	6.6250	6.6250	CIRCULAR	.0000	.0000	.0000
2	1.0000	13.2500	6.6250	6.6250	CIRCULAR	.0000	.0000	.0000
3	1.0000	13.2500	6.6250	6.6250	CIRCULAR	.0000	.0000	.0000

MAXIMUM PRIMARY PLUS SECONDARY STRESS INTENSITY

1.0000      -1.18                      .63                      -1.18                      .63                      -5.06                      3.87                      2.71                      -2.61

MAXIMUM PRIMARY PLUS SECONDARY STRESS INTENSITY

1.0000      -1.91                      1.03                      -1.91                      1.03                      -8.21                      6.30                      4.40                      -4.25

MAXIMUM PRIMARY PLUS SECONDARY STRESS INTENSITY

1.0000      -2.78                      1.56                      -2.78                      1.56                      -13.98                      10.20                      8.17                      -7.57

ORIGINATOR: C.H. O'Brien  
 PROJECT: DAVIS BRIDGE  
 UNIT 1

DATE: 12/9/84  
 JOB # 12501

CHECKER: T. Kim  
 STANDARD COMPUTER PROGRAM ME-210

DATE: 12/11/84

CALC. NUM: C-PS/PSU-1  
 REV NUM: 0

\*\*\*\*\* ME-210 POST-PROCESSOR (VERS. B) 12/8/81 \*\*\*\*\*

DO YOU WANT TO ENTER PRESSURES AND OTHER STRESSES (YES OR NO)? > YES  
 ENTER ACTUAL PIPE O.D. AND WALL (INCHES) O.D., W? > 10.75, 1.0  
 ENTER OPERATING AND DESIGN PRESSURE (PSIG) P-OP., P-DES.? > 2750, 2500  
 IS ALL DATA O.K. SO FAR (YES OR NO)? > YES  
 ENTER FOLLOWING STRESSES WT, OBE, SSE, THRM, SAM (KSI)? > .825, 4.415, 8.203, 4.546, 0.  
 ENTER SM STRESS ALLOWABLE (KSI)? > 16.48  
 IS ALL DATA O.K. (YES OR NO)? > YES

PIPE O.D. = 10.75  
 PIPE WALL = 1.000  
 DESIGN PRESSURE = 2500  
 OPERATING PRESSURE = 2750  
 WEIGHT STRESS = .825  
 OBE STRESS = 4.415  
 SSE STRESS = 8.203  
 THERMAL STRESS = 4.546  
 SAM STRESS = 0.000  
 ALLOWABLE STRESS (SM) = 16.480

CHECK	VALUE	ALLOWABLE
1A	14.03	< 24.720
1B	14.40	< 49.440
2A	12.97	< 24.720
2B	17.39	< 49.440
3A	13.08 +	< 49.440
3B	13.08 +	< 49.440

} OK

FOR CHECK 3A AND 3B YOU MUST ADD THE MAX PRI + SEC STRESS INTENSITY FROM THE ME-210 RUN FOR LOAD CASE #3 AND MUST STILL BE BELOW THE SPECIFIED ALLOWABLE



# CALCULATION COVER SHEET

Q

PROJECT Davis Besse Unit-1 JOB NO 12501009 DISCIPLINE MECH./PROCESS

SUBJECT Pipe Stress Analysis FILE NO \_\_\_\_\_

ORIGINATOR William Gallwey CALC NO W.8

CHECKER AL. Nay DATE 7/15/82 DATE 7/15/82

NO OF SHEETS 43

CALC. SHTS. = 17 ATTACHMENT #1 = 1 DWG, ATTACHMENT #2 = 4 SHTS

RECORD OF ISSUES ATTACHMENT #3 = 3 SHTS, ATTACHMENT #4 = 2 SHTS, ATTACHMENT #5 = 3 SHTS, ATTACH #6 = 2 SHTS, ATT. #7 = 10 SHTS

NO.	DESCRIPTION	BY	DATE	CHKD	DATE	APPRD	DATE	DATE FILMED
C1	ISSUED For Study	W.S.	7/15/82	AW	7/15/82	SPL	7-15-82	
C2	REVISED SHT NO. 3 AND ADDED SHT. NOS. 13, 14, 15, 16 & 17 AND ATTACH. NO. 617	SPL	12-5-84	DPO	12-5-84		12-5-84	

PRELIMINARY CALC.

COMMITTED PRELIMINARY DESIGN CALC.

SUPERSEDED CALC.

FINAL CALC.

## STUDY For FCR 82-043

ONE OF THE VERTICAL SNUBBERS FOR HANGER NO. PSU-R1 (DATA PT. 40) FAILED (NCR NO. 84-190).

IN ADDITION, THERE IS INTERFERENCE PROBLEM BETWEEN PIPE AND WALL AT THIS HANGER DUE TO THERMAL EXPANSION OF PIPE.

THIS STUDY IS MADE WITH THESE CONDITIONS.

NOTE: 1) REV. C2 IS INDEPENDENT OF REV. C1

2) REV. C2 → NCR NO. 84-190.

2.0 ISOMETRICS

The following isometrics have been used in the analysis.

Piping Iso. No.	Rev. No.	Piping Iso. No.	Rev. No.
* 1 SIM-PS/PSU	0		
⋮			

MADE FROM WALK DOWN OF SURGE LINE DESIGN DOCUMENTS

- Code  ASME Code Section III Subsection NB  
 ANSI B31.1
- System Flow Diagrams N/A Rev. \_\_\_\_\_  
 \_\_\_\_\_ Rev. \_\_\_\_\_  
 \_\_\_\_\_ Rev. \_\_\_\_\_
- Piping Class SEE 7749-M-509-1-2 Rev. \_\_\_\_\_
- Piping Class Summary ↓ Rev. \_\_\_\_\_
- SAR  Checked  Not Checked

6) MATH MODEL: ATTACHMENT 1.

7) BFW Telecopy of 7/2/82 Attachment 2.  
 " " " 6/25/82 " 3  
 " " " 7/1/82 " 4  
 COMPUTER OUTPUTS (Snumb No. & Date)

1. WEIGHT	Pipe + INSUL. (H-I active)	G 3752	6/30/82
WEIGHT	Pipe without INSUL. (H-I active)	G 3746	6/30/82
2. Weight	Pipe without INSUL. (H-I inactive)	G 3647	6/29/82
3. Seismic			
4. THERMAL (670°F)	G 7377	11-26-84	(NORMAL OPER.)
	G 7455	11-30-84	(WITH APPLIED THERMAL DISPL.) AT D.P. 41
	G 7458	12-01-84	(WITH ZERO THERMAL DISPL.) AT D.P. 41

ANALYST Siva p. King DATE 12-1-84 CHECKER DRG DATE 12-4-84

CS

GPO-13384-C 1/80



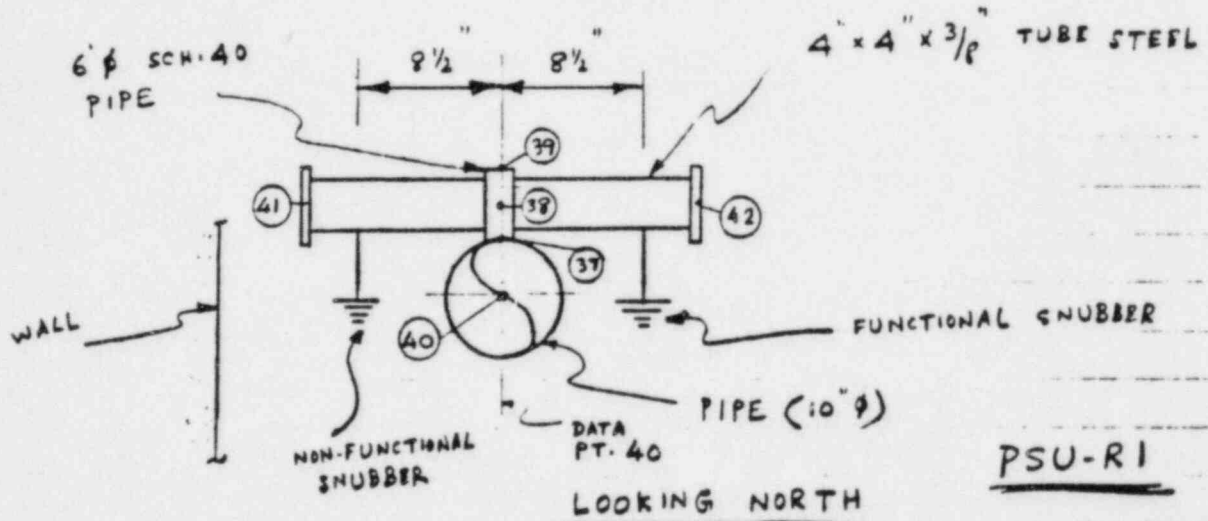




# CALCULATION SHEET

JOB NO. 12501	NCR 84-180	CALC. NO. M-8	REV NO. C2	SHEET NO. 13
PRESSURIZER SURGE LINE				
ORIGINATOR Siva P. Amgan	DATE 11-30-84	CHECKED J. J. P. Orl	DATE 12-4-84	

AS PER NCR 84-180, ONE OF THE VERTICAL SNUBBERS (CLOSE TO THE WALL) BECAME NON-FUNCTIONAL AS SHOWN.



THIS SNUBBER IS ONLY  $4\frac{1}{4}$ " AWAY FROM THE WELD OF THE NEAREST ELBOW (B&W DATA PT. 3).

THE PURPOSE OF THIS CALCULATION IS TO CHECK THE PIPE STRESSES FOR OPERABILITY CONDITION.

FROM B&W SPEC. 7749-M-505-22-6, PAGE 8.3, THE LOAD ON THIS SNUBBER IS 3900 LBS. (SSE) & 2400 (OBE).

AS PER ATTACHMENT #2 OF CALC. NO. M-8, REV. C1, (SHT. NO. 2)

THE MAX. STRESS FOR EQ. 9 OF NB-3652 OCCURS AT B&W DATA PT. 3 WHICH IS VERY CLOSE TO THIS SNUBBER.

$$\text{MAX. STRESS} = 22074.96 \text{ psi} (< 1.5 S_m = 24,800 \text{ psi @ } 670^\circ\text{F})$$

$$\text{EQ. 9} \quad B_1 \frac{P D_o}{2t} + B_2 \frac{D_o}{2I} M_i \leq 1.5 S_m$$

(ASME SPEC. III, NB-3652) 1971



# CALCULATION SHEET

JOB NO. 12501	NCR NO. 84-180	CALC. NO. M-8	REV. NO. C2	SHEET NO. 14
PRESSURIZER SURGE LINE			ORIGINATOR	DATE
			Siva P. Lingam	11-30-84
			CHECKED	DATE
			DLPolly	12-4-84

$B_1$  FOR 10" ELBOW = 1.0 (FROM TABLE NB-3623.2-1)

$$\begin{aligned}
 B_2 &= 0.75 C_2 = 0.75 \times \frac{1.95}{\left(\frac{t}{r_2}\right)^{2/3}} \\
 &= 0.75 \times \frac{1.95}{\left[\frac{(1.0) \cdot 15}{(4.875)^2}\right]^{2/3}} \\
 &= 0.75 \times 2.65 \\
 &= 1.9875
 \end{aligned}$$

$$\begin{aligned}
 D_0 &= 10.75" \\
 t &= 1.0" \\
 P &= 2500 \text{ psi} \\
 I &= 368 \text{ in}^4
 \end{aligned}$$

EQ. 9 FOR ELBOW

$$(1.0) \cdot \left( \frac{2500 \cdot 10.75}{2 \times 1.0} \right) + (1.9875) \cdot \left( \frac{10.75}{2.368} \right) \cdot M_i = 22074.96$$

$$\therefore M_i = 297542 \text{ in-lbs.}$$

THIS IS THE RESULTANT MOMENT DUE TO COMBINED WEIGHT AND SEISMIC LOADS.

FROM BECHTEL COMPUTER RUN G 3746 DATED 6-30-82,

$$\left. \begin{aligned}
 M_{xw} &= 957 \text{ FT-LBS.} \\
 M_{yw} &= 319 \text{ " } \\
 M_{zw} &= 4157 \text{ " }
 \end{aligned} \right\} \begin{aligned}
 &\text{DUE TO WEIGHT LOADS AT DATA PT. 45B.} \\
 &\text{(IGNORING THE SIGN) (EQUIVALENT TO} \\
 &\text{DLW DATAPT. 3)}
 \end{aligned}$$

$$M_i = \sqrt{(M_{xw} + M_{xs})^2 + (M_{yw} + M_{ys})^2 + (M_{zw} + M_{zs})^2}$$

$$\frac{297542}{12} = \sqrt{(957 + M_{xs})^2 + (319 + M_{ys})^2 + (4157 + M_{zs})^2}$$

ASSUME  $M_{ys} = M_{zs} = 0$  (MOMENTS DUE TO SEISMIC LOADS)

$$M_{xs} = 23485.1333 \text{ FT-LBS.}$$



# CALCULATION SHEET

JOB NO. 12501	NCR NO. 84-128	ALC. NO. M-8	REV. NO. C2	SHEET NO. 15
PRESSURIZER SURGE LINE				
ORIGINATOR Swa P. Kingam	DATE 11-30-84	CHECKED D. J. POBY	DATE 12-4-84	

SINCE ONLY ONE SNUBBER IS FUNCTIONAL, THE PIPE WILL EXPERIENCE AN ADDITIONAL MOMENT,  $M_x(N-S)$  DUE TO  $8\frac{1}{2}$ " ARM LENGTH.

$$M_x = 2400 \times 8.5 = 20400 \text{ in-lbs. (OBE)}$$

$$= 3900 \times 8.5 = 33150 \text{ in-lbs. (SSE)}$$

ADDING THIS,

$$M_{XS} = (23485.1333) \times 12 + 20400 = 302222 \text{ in-lbs. (OBE)}$$

$$= (1.975)(23485.1333) \times 12 + 33150 = 561566 \text{ in-lbs. (SSE)}$$

$$M_{YS} = M_{ZS} = 0$$

$$M_i = \sqrt{(957 \times 12 + 302222)^2 + (319 \times 12 + 0)^2 + (4157 \times 12 + 0)^2}$$

$$= 317671 \text{ in-lbs. (OBE)}$$

$$= 575230 \text{ in-lbs. (SSE)}$$

EQ. 9 FOR ELBOW WITH NON-FUNCTIONAL SNUBBER (ONE)

$$(1.0) \times \left( \frac{2500 \times 10.75}{2 \times 1.0} \right) + (1.9875) \times \left( \frac{10.75}{2 \times 368} \right) \times 317671 = 22,660 \text{ psi}$$

( $< 1.55 S_m = 24,800 \text{ psi}$ )

OBE  $\rightarrow$  UPSET CONDITION

$$(1.0) \times \left( \frac{2500 \times 10.75}{2 \times 1.0} \right) + (1.9875) \times \left( \frac{10.75}{2 \times 368} \right) \times 575230$$

$$= 30,137 \text{ psi}$$

( $< 3 S_m = 49,600 \text{ psi}$ )

SSE  $\rightarrow$  FAULTED CONDITION



# CALCULATION SHEET

JOB NO. 12501 NCR NO. 84-180 CALC. NO. M-8		REV. NO. C2	SHEET NO. 16
PRESSURIZER SURGE LINE			
ORIGINATOR Siva P. Arangan	DATE 11-30-84	CHECKED D. J. O'Neil	DATE 12-4-84

$$M_i \text{ due to seismic} = 302222 \text{ in-lbs. (OBE)}$$

$$561566 \text{ in-lbs. (SSE)}$$

$$\sigma_{OBE} = \frac{D_o}{2I} \times M_i = \left( \frac{10.75}{2 \times 368} \right) \times 302222 = 4415 \text{ psi (OBE)}$$

$$\sigma_{SSE} = \left( \frac{10.75}{2 \times 368} \right) \times 561566 = 8203 \text{ psi (SSE)}$$

$$\sigma_{WT.} = 825 \text{ psi (weight)} \rightarrow G 3746 \text{ dated 6-30-82}$$

$$\sigma_{TH.} = 4546 \text{ psi (thermal)} \rightarrow G 7377 \text{ dated 11-26-84}$$

Also, the cumulative usage factor at B&W data pt. 3 is only 0.045 which is  $\ll 1.0$ .

## conclusions

The stresses are within allowables even if one of the snubbers is non-functional.

Therefore, the operability of the plant is not affected due to pipe stresses.

Since the primary faulted stresses are approximately 60% of the allowable, a slight shift in frequency due to non-functional snubbers will not significantly affect the results.





# CALCULATION SHEET

JOB NO. 12501	NCR NO. 84-180	CALC. NO. M-8	REV. NO. C2	SHEET NO. 17
PRESSURIZER SURGE LINE				
ORIGINATOR Siva P. Kungam	DATE 12-4-84	CHECKED [Signature]	DATE 12-4-84	

## THERMAL EVALUATION DUE TO INSUFFICIENT CLEARANCE AT WALL

AS PER ATTACHMENT NO. 6, THERMAL RUNS ARE MADE WITH AND WITHOUT APPLIED THERMAL DISPLACEMENTS AT DATA PT. 41.

THE PIPE STRESSES ARE WITHIN ALLOWABLES FOR BOTH CLASS-1 AND CLASS-2

$$\text{CLASS 1} \rightarrow C_1 \cdot \frac{D_o}{2I} M_i \leq 3 S_m$$

$$C_1 = 2.65$$

$$\text{CLASS 2} \rightarrow i \cdot \frac{M_c}{2} \leq S_A$$

HOWEVER, FOR LONG TERM OPERATION, THE PIPE MUST BE EVALUATED FOR FATIGUE (CUMULATIVE USAGE FACTOR  $\leq 1.0$ ).

FOR SHORT TERM OPERABILITY, THE PIPE IS SAFE FOR FATIGUE BECAUSE OF UNUSED CYCLES.

BY ENGINEERING JUDGEMENT, THE LOADS AT PRESSURIZER AND HOT LEG ARE REASONABLE FOR OPERABILITY.



MEMORANDUM

"DON'T SAY IT - WRITE IT"

To SIVA LINGAM Location \_\_\_\_\_  
 From MARC NEMSER Date 12/3/84  
 Subject: NCR 84-0180, FAILED SNUBBER PSU-R1

Based on the information I have received from the reanalysis of the Pressurizer Surge line (Thermal movements 1.651" north, 0.177" up, & 0.69" west at data points 40 and 35), I have projected the movements on a scaled drawing to determine interferences. The drawing, copy attached, was prepared for this comparison and uses dimensions from the original Grinnell support design drawing, as well as from the actual dimensions measured in the plant. Because of the possibility of some dimensional inaccuracies, I would suggest that the reanalysis be performed using 3 different interference conditions. The initial run should be calculated considering full restraint (interference) at  $\Delta Y = 0.125$ " up and  $\Delta Z = 0.375$ " west. This is conservative, but most accurately reflects the actual interference that occurred. Then, for comparison purposes, run the problem using interference at  $\Delta Y = 0.0625$ " up and  $\Delta Z = 0.25$ " west, and finally with no permissible movement,  $\Delta Y = 0$ ,  $\Delta Z = 0$ . The results of this analysis should help us to determine the operability of this pipe.

ATTACHMENT NO. 6

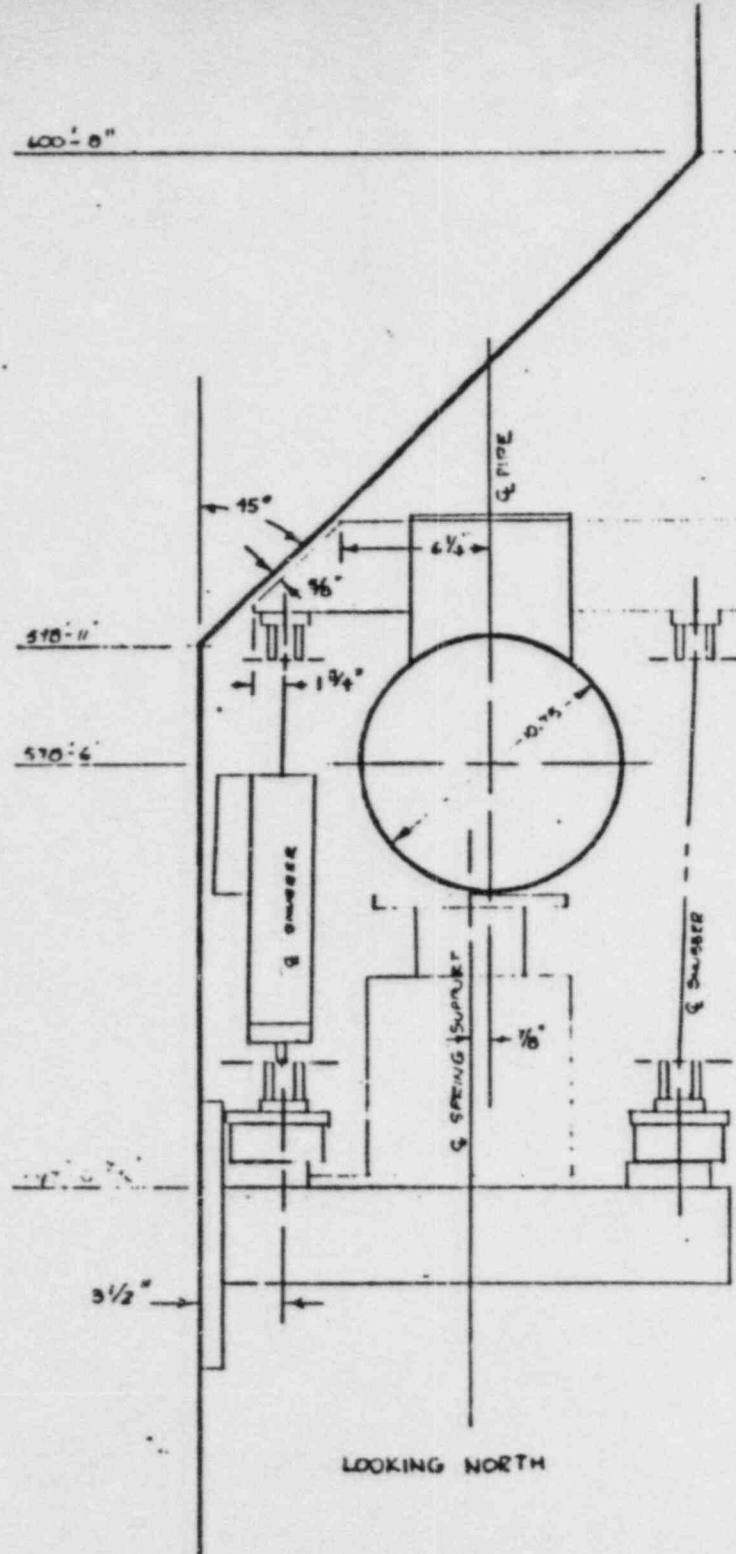
SHT. NO. 1 OF 2

PROB. NO. M-8, REV. C2

*M. Nemser*

Added by: Siva P. Lingam date: 12-4-84

ch'd by: JOP ORG date: 12-4-84



ATTACHMENT NO. 6

SHT. NO. 2 OF 2

PROB. NO. M-8, REV. C2

Added by: Siva P. Lingam Date: 12-4-84

Ch'kd by: D. J. [Signature] Date: 12-4-84

SKETCHED BY: M. Venner Date: 12-3-84

PSU-R1  
SCALE: 1/8" = 1"

Bechtel Associates  
Professional Corporation (Ohio)

15740 Shady Grove Road  
Gaithersburg, Maryland 20877-1454  
301-258-3000



TELECOPY

Mr. J. K. Wood  
General Supervisor  
Facility Engineering  
The Toledo Edison Company  
P. O. Box 929  
Toledo, Ohio 43652

DEC 5 1984

Dear Mr. Wood:

The Toledo Edison Company  
Davis-Besse Nuclear Power Station  
Bechtel Job 12501  
TED NCR 84-180  
HYDRAULIC SNUBBER PSU-R1  
File: 0270, 1523D, 1470, 283293  
BT-15086 (A218)

Attached please find our recommended disposition of TED NCR 84-180 telecopied on November 7, 1984. This NCR addresses seismic restraint PSU-R1 on the Pressurizer Surge line, found with one (1) broken hydraulic snubber. Our evaluation and reanalysis of this condition has identified that a thermal interference existed during normal plant operating conditions. This interference resulted from insufficient clearance between the restraint and the secondary shield wall.

All stress levels on the Pressurizer Surge line and supports were found to be within interim criteria for both the condition of the actual thermal interference and a postulated SSE event. However, for long term plant operation the restraint must be modified in accordance with the attached sketch.

In addition, B&W specification 3002/NSS-14/1077 page 8.6 (7749-M-505-22-6) should be revised to show the correct thermal displacements, presently shown reversed between node points 1 and 2. This change has been incorporated in our sketch of the restraint modification. In addition to the revised movements, a new cold spring setting for PSU-H1, and revised hot and cold piston settings for PSU-R1 have also been included.

As discussed during the engineering review meeting held here on November 28, 1984, we will require, in addition to the 115 jobhours already authorized, 130 jobhours to account for our engineering effort to design the restraint modification and complete the operability evaluation.

Also as discussed, it will require 200 jobhours for us to amend B&W's Nuclear Class 1 stress report for this line to reflect the modified support configuration and the effect of the thermal interference.

Added by: Siva P. Arinam 12-5-84

ch'ed by: [Signature] 12-5-84

ATTACHMENT NO. 7

SHT. NO. 1 OF 10

PROB. NO. M-8, REV. C2



DEC 5 1984.

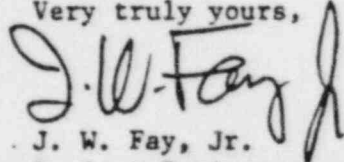
Mr. J. K. Wood

2

BT-15086

If you have any questions, please contact us.

Very truly yours,



J. W. Fay, Jr.  
Project Engineer

JWF/RDK/jil

Attachment: As stated above

cc: J. F. Helle w/o

C. T. Daft w/l

R. Rosenthal w/o

Records Management - Mail Stop 3021 w/l

ATTACHMENT NO. 7

SHT. NO. 2 OF 10

PROB. NO. M-8, REV. C2

Added by: Siva P. Singam Date: 12-5-84

ch'ed by: D. Leary Date: 12-5-84

NOV 7 1984

ATTACHMENT NO. 7

SHT. NO. 3 OF 10

PROB. NO. M-8, REV. C2

Added by: Siva P. Aringam date: 12-5-84

Checked by: [Signature] date: 12-5-84



JAVIS-BESSE NUCLEAR POWER STATION - UNIT NO. 1  
NONCONFORMANCE REPORT

DATE (8)	ORIGINATOR (9)	REPORT NO. (1)	PAGE (2)
5 Nov 1984	John L. Schmitt	84-0180	1 OF
DCS SIGNATURE (6)	DATE (10)	HOLD TAG(S) NO. (4)	INVOICER NO. (3)
[Signature]	11/5/84	0	1-84-1089-00
	IDENTITY NO. (11)	PURCHASE ORDER NO. (7)	CONDITIONAL RELEASE (12)
	1.2153	N/A	N/A

SUBJECT (13)  
HYDRAULIC SNUBBER PSU-R1 (D-B No. C-203)

NONCONFORMANCE DESCRIPTION (13)  
DURING THE VISUAL INSPECTION PROGRAM FOR 1984 PER SECTION 4.1 OF ST 5014.01,

THE ABOVE SUBJECT, LOCATED NEXT TO CONCRETE WALL, PISTON ROD WAS FOUND  
DISMEMBERED TOTALLY - IE: SNAPPED INTO. A REVIEW OF RECORDS  
SHOW THAT SUBJECT WAS VISUALLY INSPECTED 7 AUG 1983 TO SATISFY

GROUP (14)	CONTRACTOR'S NAME	CAUSE (15)
<input type="checkbox"/> MAINT. ELECT <input type="checkbox"/> MAINT. I & C <input type="checkbox"/> MAINT. PIPE <input type="checkbox"/> MAINT. WELD <input type="checkbox"/> OTHER	<input type="checkbox"/> CONTRACTOR ELECTRICAL <input type="checkbox"/> CONTRACTOR FITTER <input type="checkbox"/> CONTRACTOR WELD <input type="checkbox"/> CONTRACTOR OTHER <input type="checkbox"/> NOT APPLICABLE	<input type="checkbox"/> INADEQUATE PROCEDURE <input type="checkbox"/> PROCEDURE IMPLEMENTATION <input type="checkbox"/> WORKMANSHIP <input type="checkbox"/> PERSONNEL ERROR <input type="checkbox"/> EQUIPMENT RELATED <input type="checkbox"/> WRONG MATERIAL <input type="checkbox"/> DOCUMENTATION <input checked="" type="checkbox"/> OTHER P/A STEEL-64 <input type="checkbox"/> NOT APPLICABLE
RESPONSIBLE PARTY FOR INDICATING DISPOSITION (16)	<input checked="" type="checkbox"/> FACILITY ENGINEERING <input type="checkbox"/> MAINTENANCE ENGINEER	<input type="checkbox"/> FACILITY MODIFICATION <input type="checkbox"/> PURCHASING <input type="checkbox"/> OTHER

DISPOSITION (17)  
 USE-AS-IS     USE-AS-IS TEMPORARILY UNTIL (Date)     REPAIR     REWORK     RESET

DISPOSITION JUSTIFICATION (17)  
 Rework seismic restraint PSU-R1; Thermal movement interference between the west hydraulic snubber (part 4) and the secondary shield wall, caused the piston shaft to break at the upper end paddle. Also, continuing thermal movement caused interference between the upper (cont on page 5)

RESPONSIBLE PARTY FOR IMPLEMENTATION OF DISPOSITION (18)  
 FACILITY ENGINEERING     MAINTENANCE ENGINEER     PURCHASING     OTHER

ENGINEERING CONCURRENCE OF DISPOSITION (Use As-Is, Use-As-Is Temporarily, Repair Only) (21)

CAN ITEM PERFORM ITS INTENDED SAFETY FUNCTION (22)  
 YES     NO    If NO, Fully Tested System

SIGNATURE (Facility Engineering General Supervisor) (23)    DATE (24)

IMPLEMENTATION OF DISPOSITION (25)

SIGNATURE (26)    INVOICER NO. (28)    SUPPLYER NO. (29)    DATE (27)

DISTRIBUTION (30)	FINAL ACCEPTANCE
<input checked="" type="checkbox"/> QC NON FILE <input checked="" type="checkbox"/> STATION SUPERINTENDENT <input type="checkbox"/> MAINTENANCE <input checked="" type="checkbox"/> NUCLEAR FACILITY ENGINEERING <input checked="" type="checkbox"/> CODE INSPECTION/AM <input type="checkbox"/> FACILITY MODIFICATION	GD INSPECTOR (31)    DATE (32) HOLD TAGS REMOVED (Signed) (31)    DATE (33) AUTHORIZED NUCLEAR INSPECTOR (Code Name Only)    DATE (34)

COMPLETED BY THIS DC

COMPLETED BY RESPONSIBLE PARTY FOR INDICATING DISPOSITION

COMPLETED BY FACILITY ENGINEERING

COMPLETED BY QUALITY CONTROL

COMPLETED BY THIS DC



NONCONFORMANCE REPORT

ED 8750.1

SUBJECT

HYDRAULIC SNUBBER PLW-21

PAGE	2 OF
NCR NO.	84-0180

CONTINUATION FROM PAGE 1

THE ST. WITH NO FINDING RECORDED. A REVIEW OF RECORDS FOR  
 INSTALLATION INSPECTION WAS PERFORMED. INSTALLATION OF SNUBBER  
 WAS IN 1982 WITH INSPECTION BEING PERFORMED 12 JULY 1982. PISTON  
 SETTING WAS RECORDED AT 3 7/16" WHICH WAS WITHIN CRITERIA OF DUG  
 M-618. FUNCTIONAL TEST WAS ACCOMPLISHED ON 5 MAY 1982 WITH  
 TENSION LOCK UP 7.86"/MIN. TENSION BLEED RATE 5.1"/MIN.  
 COMPRESSION LOCK UP 8.23"/MIN. COMPRESSION BLEED RATE 5.2"/MIN.  
 THESE WERE WITHIN ACCEPTABLE CRITERIA FOR GROUP III.

NCRD IS REQUESTED TO EVALUATE (NCR 80-871 PREVIOUSLY  
 WRITTEN AGAINST THIS OVERALL NCR.) THE ABOVE AND PROVIDE  
 DISPOSITION.

ATTACHMENT NO. 7

SHT. NO. 4 OF 10

PROB. NO. M-8, REV. C2

Added by: Siva P. Kingam Date: 12-5-84

ch'ed by: D. POPEL Date: 12-5-84

Disposition Justification (17) (cont)

cross member (part 7) and the secondary shield wall at the 45° vertical offset. All stress levels resulting from reanalysis of the pipe and pipe supports on this are within interim criteria. This reanalysis considers the effects of both thermal interference with the wall and one (1) snubber inactive.

The support should be modified in accordance with the attached sketch, 1) deleting the west 1-1/2" snubber; 2) replacing the east 1-1/2" snubber with a 2-1/2"; 3) cutting the upper cross member to eliminate the interference; 4) adding a vertical brace.

ATTACHMENT NO. 7  
SHT. NO. 5 OF 10  
PROB. NO. M-8, REV. C2

Added by: Siva P. Kingam Date: 12-5-84

ch'ed by: D. D. P. R. S. Date: 12-5-84



TOLEDO EVISON CO.

CHECK BY: RE-DRAWN BY: DATE: 3-16-76

ORDER OR CONT. NO. 7749-11-190 UNIT #1  
JOB NAME

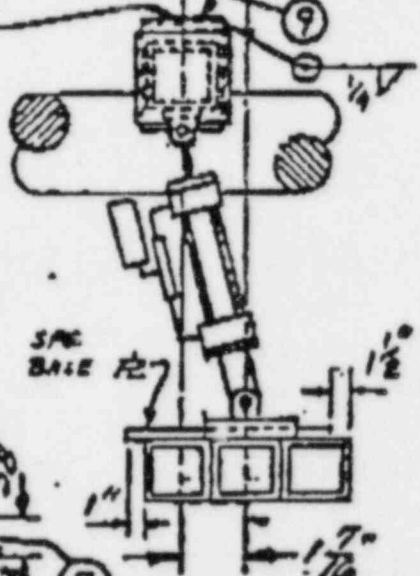
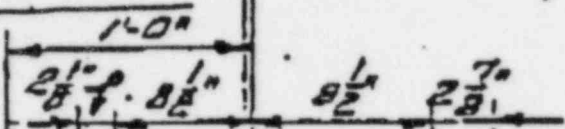
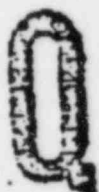
REV.	1	2	3	4	5
DATE					

ATTACHMENT NO. 7  
SHT. NO. 6 OF 10

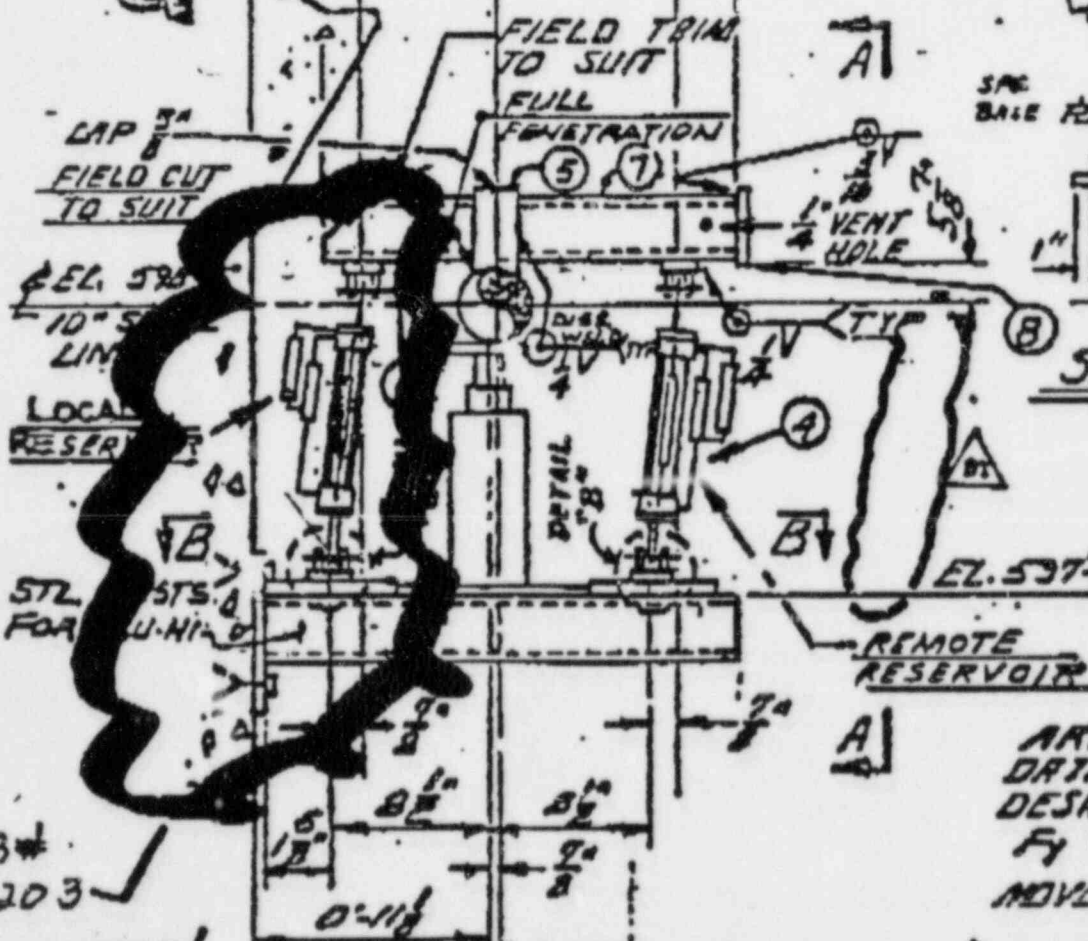
Prov. Hqr. Dept. Approved By: Date: 5/17/77

PROB. NO. M-8, REV. C2

TOP BRACKET & BOTTOM BRACKET



SECTION A-A



RECEIVED JUN 07 1977 POWER ENGR.

AREA #3  
DATA PT. #3  
DESIGN LOAD  
F<sub>y</sub> = 4485  
MOVEMENTS

Δx = 1 7/8"  
Δy = 4 1/2"  
Δz = 1 7/8"

171214

NOTE!

WORK THIS SKETCH WITH NOTE!  
B/W SKETCH NO. 202 INSTALLATION  
HANGER PSU-NI BY OTHERS  
7749-11-190-HBW-203-~~X~~ HW-203A F2 AND  
FOR MATERIALS AND OPERATIONS SEE SKETCH NO. 202

REF. DWG. NO. PIPE: 21-271-7 OVER: C-124-7 MAKE NO. SKETCH NO. HW-203 F2 AND

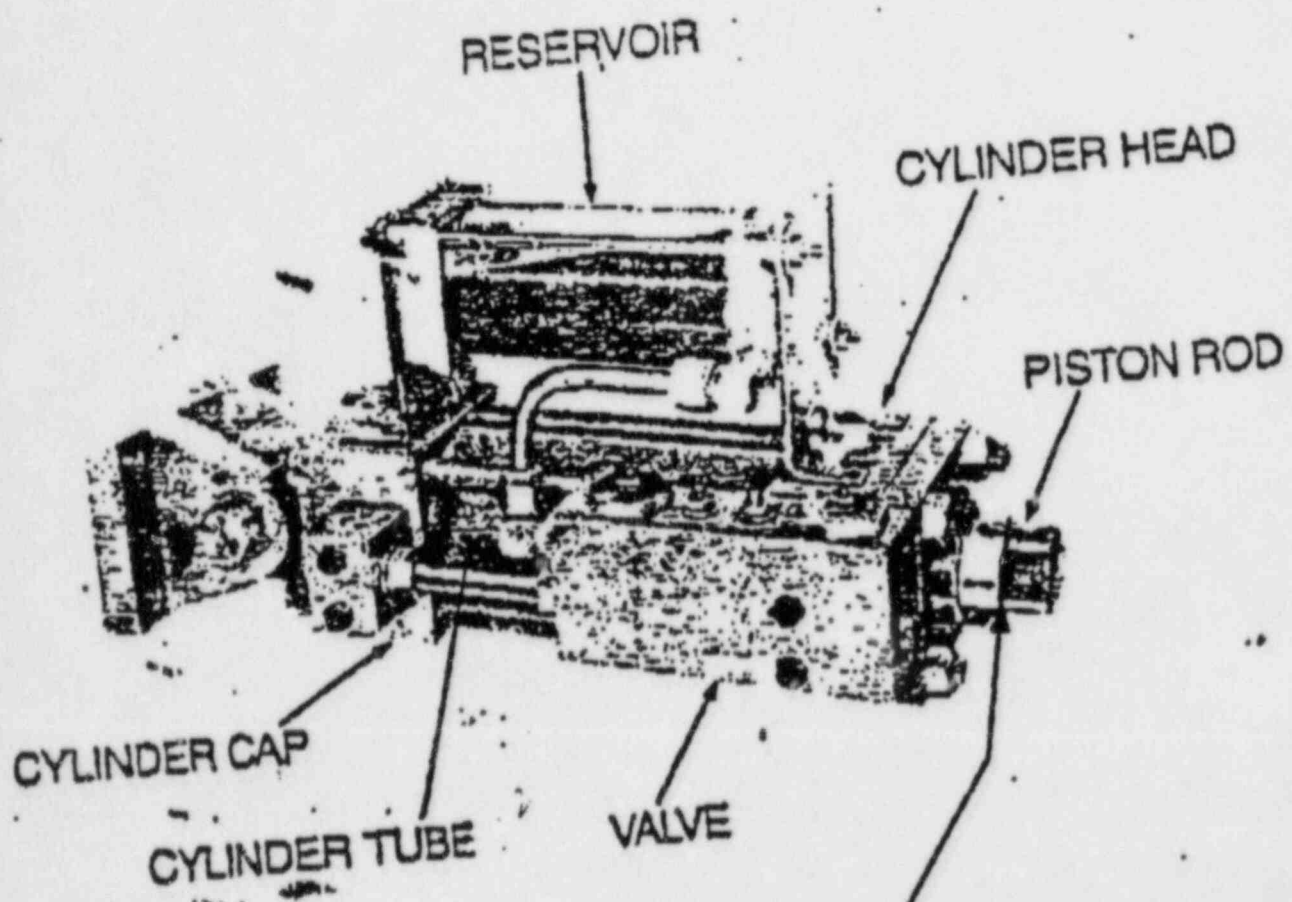
Added by: Siva P. Lingam date: 11-5-84

ch'ed by: J. P. ORY date: 12-5-84

Quality Control  
DOCUMENT ATTACHMENT  
NCR 84-0180  
PAGE 3 OF 3

Hydraulic Snubber  
Basic components of all hydraulic snubbers are:

Quality Control  
DOCUMENT ATTACHMENT  
NCR 84-0180  
PAGE 4 OF



LOCATION OF PISTON ROD  
DISMEMBERMENT

ATTACHMENT # 7  
SHT. NO. 7 OF 10  
PROB. NO. M-P, REV. C2

Added by: Siva P. Arigam date: 12-5-84

Checked by: D. P. O'Kelly date: 12-5-84

ATTACHMENT NO. 7

SHT. NO. 8 OF 10

PROB. NO. M-8, REV. C2

Attachment to BT-15086

Added by: Siva P. Kingam 12-5-84

Checked by: [Signature] 12-5-84

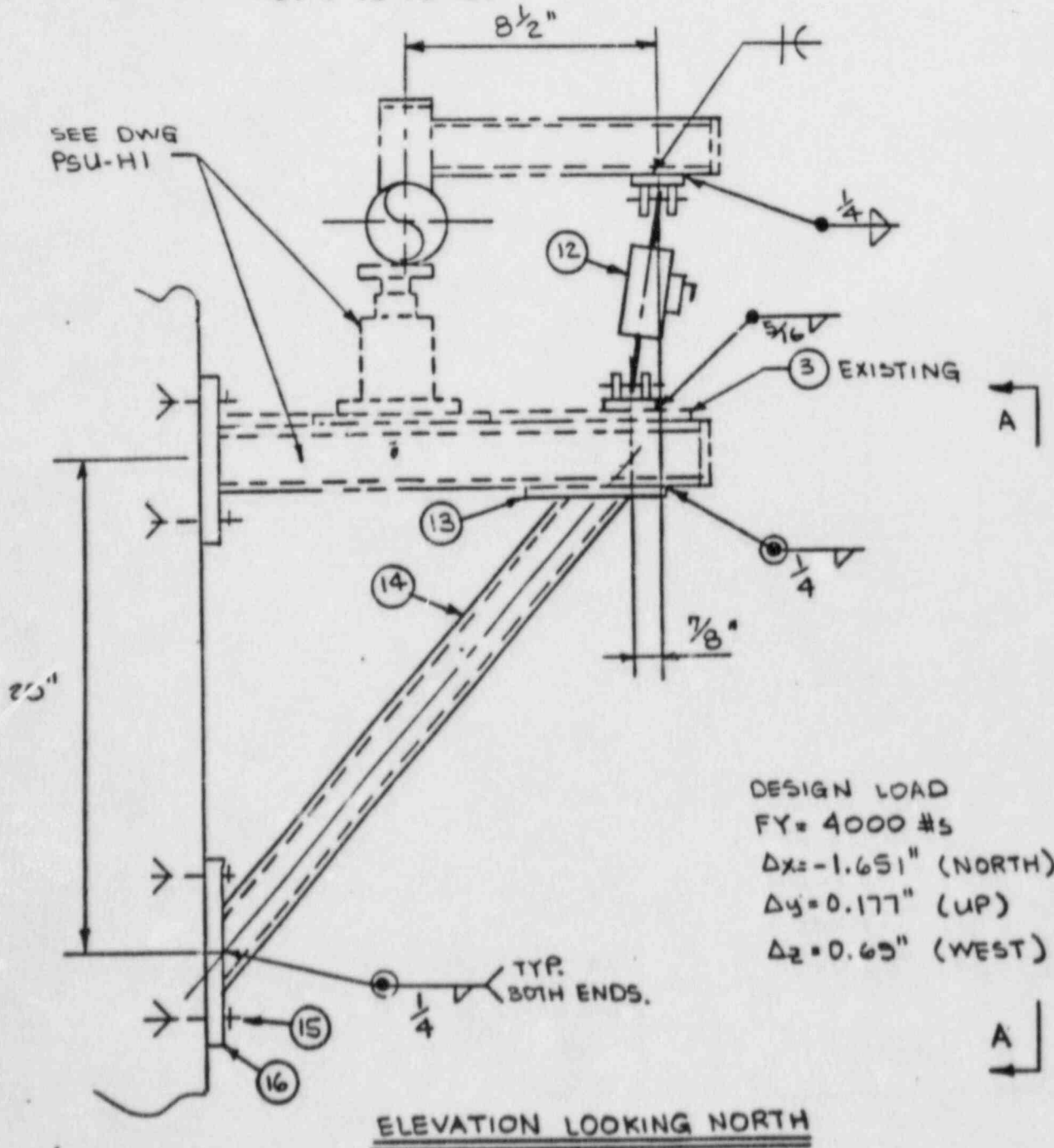
DWG. PSU-R1 REV. B2

Q

SHEET 1 OF 3

DESCRIPTION OF CHANGE:

1. REMOVE EXISTING ITEMS 4, 6, 10 AND 11.
2. MODIFY SUPPORT AS-SHOWN.
3. WORK THIS SUPPORT WITH DWG. PSU-H1.
4. ADD NEW ITEMS 12 TO 16.



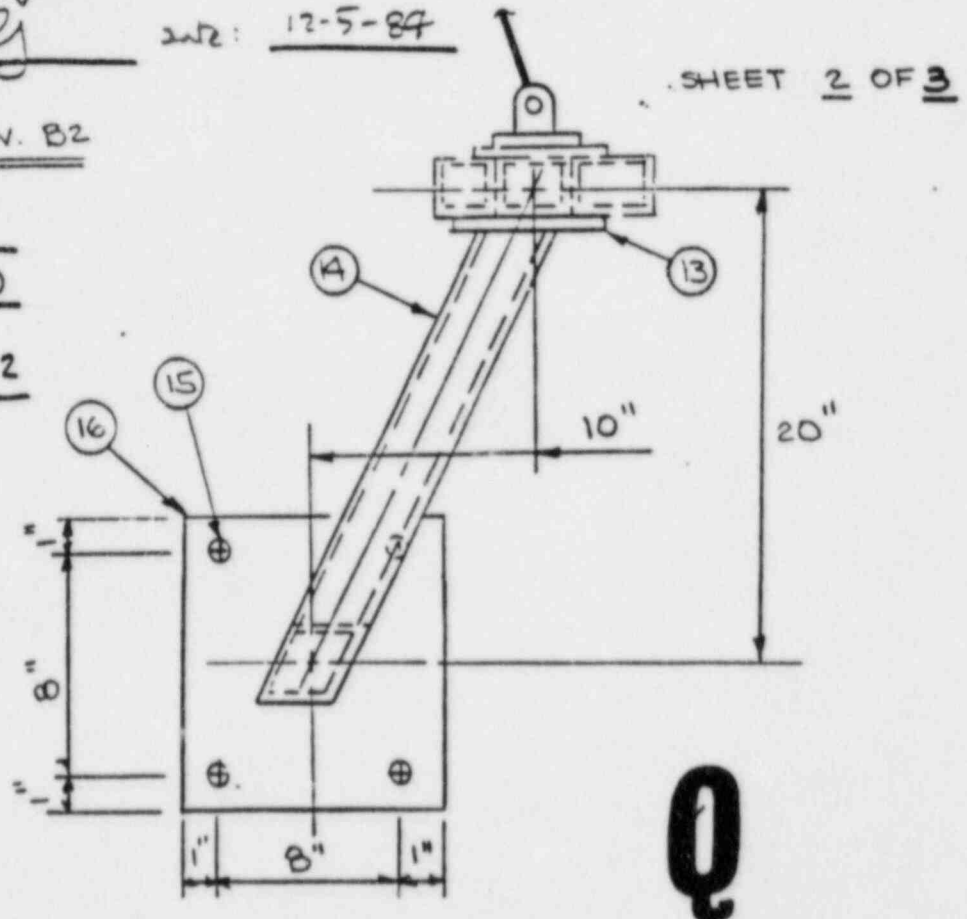
Added by: Siva P. Kingam Date: 12-5-84  
 checked by: REPORLY Date: 12-5-84

DWG. PSU-R1 REV. B2

ATTACHMENT NO. 7

SHT. NO. 9 OF 10

PROB. NO. M-8, REV. C2



SECTION "A-A"

ITEM	QTY	DESCRIPTION	MATERIAL
12	1	2 1/2" CYLINDER 5" STROKE FIG. 200 OPTION 2 HYDRAULIC SHOCK & SWAY SUPPRESSOR	
13	1	1/2" X 8" X 8" PLATE	ASTM A-36
14	1	TS 4 X 4 X 3/8 X 3'-0" (FIELD COPE TO SUIT)	ASTM A-500 GR. B
15	4	1/2" X 3 3/4" HILTI KWIK BOLT MIN. EMB = 2 1/4"	ASTM A-108 GR. 1144
16	1	1/2" X 10" X 10" PLATE W/(4) 1/4" HOLES	ASTM A-36

NOTES:

1. FABRICATE & INSTALL PER SPEC. 12501-M-450Q.
2. FOR ACCEPTABLE RANGES OF HOT AND COLD PISTON SETTINGS REFERENCE DWG. 12501-M-618. \*
3. PAINT AS REQ'D PER SPEC. 12501-A-24Q.

\* REVISED RANGE OF HOT & COLD PISTON SETTINGS  
 TABLE 1 PAGE 6 OF 6

HL-PS-PSU	PSU-R1	CPS (MIN)	CPS (MAX)	HPS (MIN)	HPS (MAX)	LV/DR GROUP
		1-5/8"	5-1/2"	1-1/8"	5-3/8"	III



DWG. PSU-HI REV. B2

Q

SHEET 3 OF 3

RESET SPRING TO:

HL = 4282 #

CL = 4565 #

REVISE HANGER DWG. TO INCLUDE:

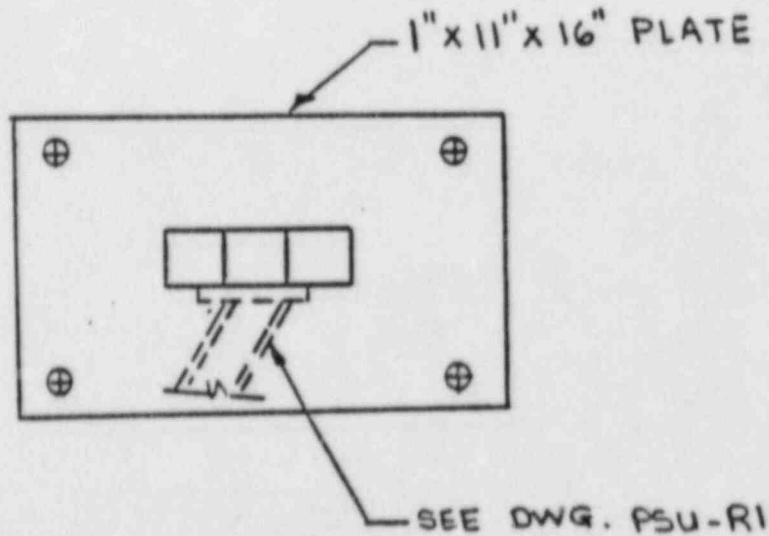
DESIGN LOAD

FY = 4565 #

$\Delta x = -1.651''$  (NORTH)

$\Delta y = 0.177''$  (UP)

$\Delta z = 0.69''$  (WEST)



SECTION "A-A"

ATTACHMENT NO. 7

Added by: Siva P. Kingam date: 12-5-84

SHT. NO. 10 OF 10

Checked by: Dal Röhl date: 12-5-84

PROB. NO. M-8, REV. C2



# CALCULATION COVER SHEET

Q

PROJECT Davis Besse Unit-1 JOB NO 12501009 DISCIPLINE P.D.E.  
 SUBJECT Pipe Stress Analysis FILE NO \_\_\_\_\_  
 ORIGINATOR William Gallagher CALC NO m8  
 CHECKER AL. Nay DATE 7/15/82 DATE 7/15/82  
 NO OF SHEETS 26

CALC. SHTS. = 12. ATTACHMENT # 1 = 1 DWG, ATTACHMENT # 2 = 4 SHTS

RECORD OF ISSUES ATTACHMENT # 3 = 3 SHTS. ATTACHMENT # 4 = 2 SHTS  
 ATTACHMENT # 5 = 3 SHTS.

NO	DESCRIPTION	BY	DATE	CHKD	DATE	APPRD	DATE	DATE FILMED
1	ISSUED For Study	WFG	7/15/82	ALW	7/15/82	SPL	7-15-82	
△								
△								
△								
△								
△								

PRELIMINARY CALC.   
 SUPERSEDED CALC.

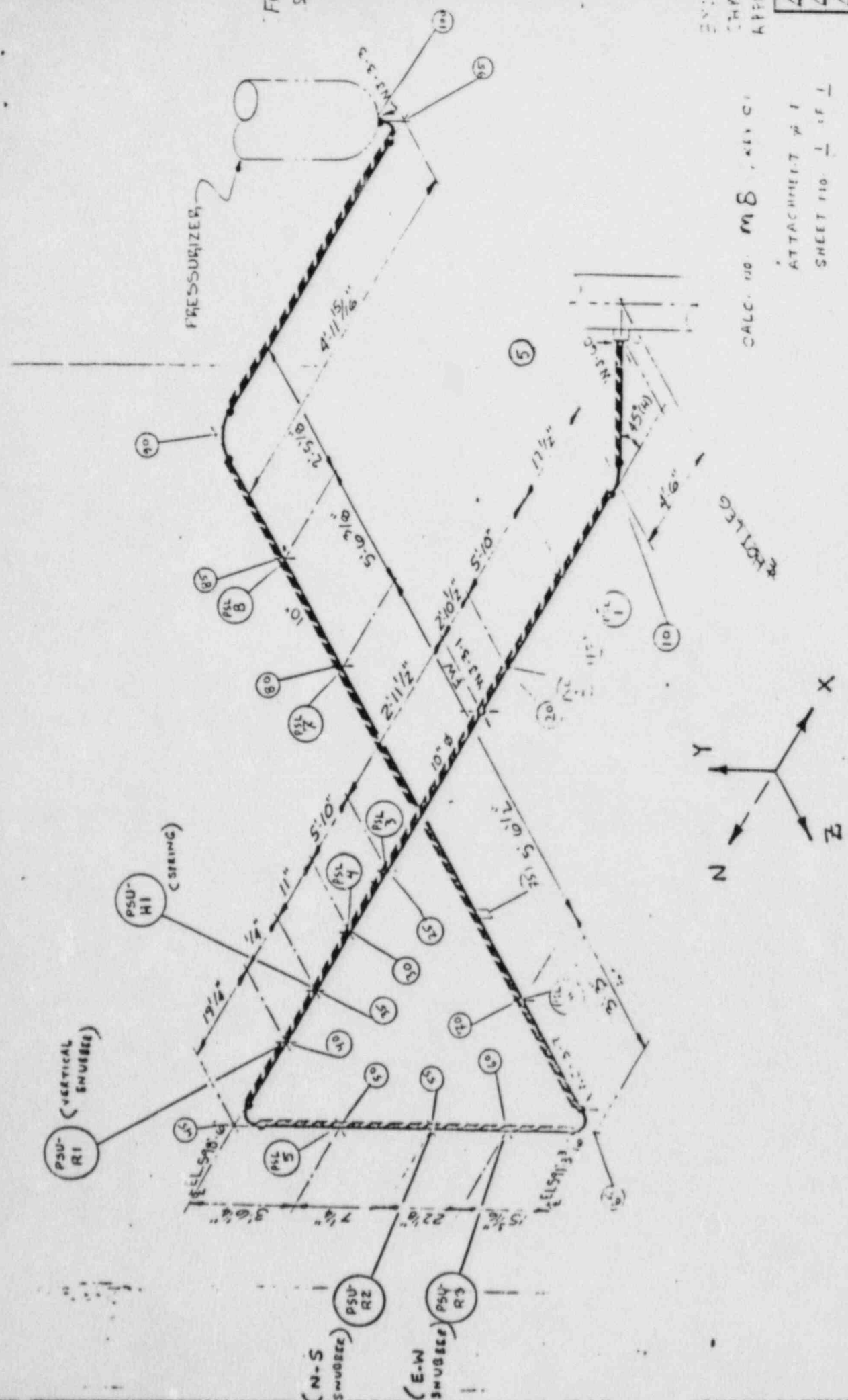
COMMITTED PRELIMINARY DESIGN CALC.   
 FINAL CALC.

STUDY For FCR 82-043

FO 58

EV: CHED APP

CALC. NO. M8, REV. C1  
ATTACHMENT # 1  
SHEET 110 OF 1



pressurizer surge line

Introduction

This report contains the stress analysis of the surge line. All points in the surge line are analyzed including the bimetallic weld at the hot leg surge nozzle. The analysis of the intersection at the pressurizer end is contained in pressurizer Design Report "Surge Nozzle Analysis". The branch intersection analysis is contained in this Design Report under a different section "Surge Nozzle".

Results

All points in the surge line satisfy primary stress limits either in accordance with equation 9 or per Appendix F of the B31.7 code.

Per equation 9 simplified analysis

- Maximum Primary Stress = 22074.96 psi < 1.5 S<sub>m</sub> = 24800. psi @ 670°F
- Maximum for an elbow (joint 3) = 22074.96 psi < 24800. psi
- Maximum for a straight (joint 1) = 10220.54 psi < 24800. psi

All points in the surge line do not satisfy the 3 S<sub>m</sub> primary plus secondary stress limit. However, calculations are provided showing that an Elastic-Plastic Analysis is applicable.

- Maximum primary plus secondary stress (Bimetallic Weld) (joint 54) = 75997.62 psi > 3 S<sub>m</sub> = 50100. psi @ 650.°F (PAGE C-11)
- Maximum primary plus secondary stress for an elbow (joint 2) = 67368.71 psi > 50100. psi (PAGE C-13)
- Primary plus secondary stress for cycles occurring more than 240 times:
  - Joint 2 = 36860.4 psi < 3 S<sub>m</sub> = 50100. psi
  - Joint 3 = 33787.9 psi < 3 S<sub>m</sub> = 50100. psi
  - Joint 19 = 36929.7 psi < 3 S<sub>m</sub> = 50100. psi

Maximum Usage Factors:

- Usage Factor at Bimetallic weld (joint 2) = 0.87 < 1.0 = allowable
- Usage Factor for joint 3 = 0.045 < 1.0 = allowable
- Usage Factor for joint 19 = 0.036 < 1.0 = allowable

ALC. # m 8

REV. C1

Added by: W. F. S. I.

Date: 7/15/82

Checked by: AL. NAY

Date: 7/15/82

Ref: "Stress Analysis of Surge Line, Report #6, for Toledo Edison Company, Davis-Besse," 620-0014-50, Jun 1972 (Rev 1, 5/3/74). RP 7/2/82

ATTACHMENT # 2

SHEET NO. 2 OF 4



Randy Kies ATTN: SIVA LINGAM

M E C C I )  
SH 20F3

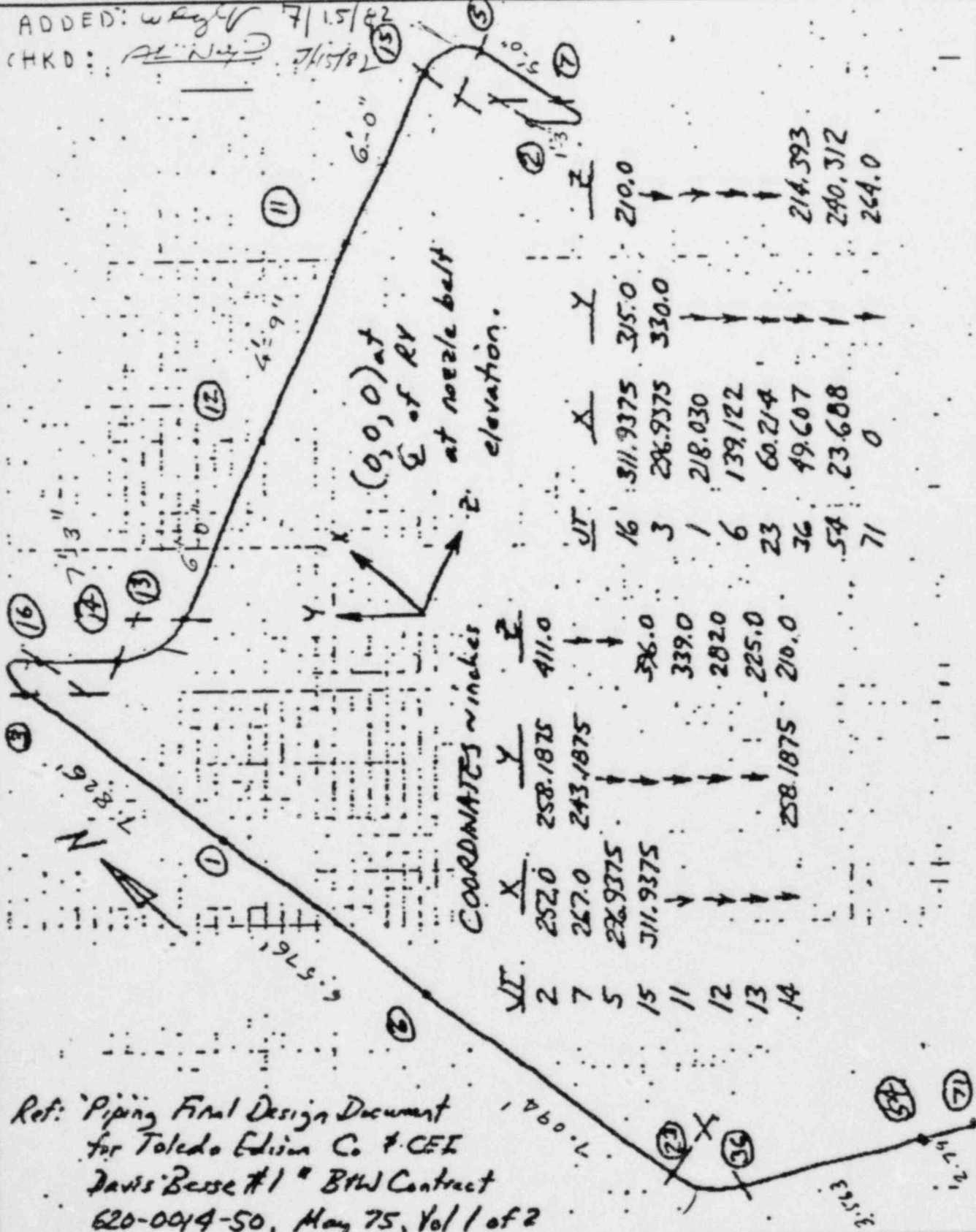
**Babcock & Wilcox**  
Nuclear Power Generation Division

ATT.# 3

PGS. 71027.1 P.23

**GENERAL CALCULATIONS**

ADDED: wright 7/15/82  
CHKD: R. N. 7/15/82



ST.	X	Y	Z	ST.	X	Y	Z
2	252.0	258.1875	411.0	16	311.9375	315.0	210.0
7	267.0	243.1875	386.0	3	286.9375	330.0	214.393
5	286.9375		339.0	1	218.030		240.312
15	311.9375		282.0	6	139.122		264.0
11			225.0	23	60.214		
12			210.0	36	49.607		
13				54	23.688		
14				71	0		

Ref: Piping Final Design Document  
for Toledo Edison Co. & CEI  
Davis-Besse #1 - BW Contract  
620-0014-50, May 75, Vol 1 of 2

PREPARED BY RP Potkin DATE 6/25/82  
REVIEWED BY \_\_\_\_\_ DATE \_\_\_\_\_

DOC. NO. \_\_\_\_\_  
PAGE NO. \_\_\_\_\_

Randy Kies

ATTN: SIVA LINGAM

M 8 (C 1)  
S.H. 3. F 3

Babcock & Wilcox

ATTCH. # 3

DOC. 21037.1 P. 49

Nuclear Power Generation Division

### GENERAL CALCULATIONS

#### DAVIS-BESSE SURGE PIPING THERMAL DISPLACEMENTS

JT	87% PWR			159% PWR		
	DX	DY	DZ	DX	DY	DZ
12	2.93E-3	-8.67E-1	1.76E-3	2.93E-3	-8.67E-1	1.71E-3
7	8.50E-2	-9.60E-1	1.77E-2	8.50E-2	-9.60E-1	1.74E-2
5	2.56E-1	-9.88E-1	8.95E-2	2.56E-1	-9.88E-1	8.89E-2
15	3.98E-1	-9.84E-1	5.40E-2	3.98E-1	-9.88E-1	5.32E-2
11	6.56E-1	-8.72E-1	-2.72E-1	6.55E-1	-8.67E-1	-2.73E-1
12	9.22E-1	-7.15E-1	-5.98E-1	9.19E-1	-7.03E-1	-5.99E-1
13	1.14E0	-5.27E-1	-9.24E-1	1.14E0	-5.09E-1	-9.25E-1
14	1.26E0	-3.68E-1	-9.54E-1	1.26E0	-3.67E-1	-9.54E-1
16	1.59E0	-6.10E-2	-7.41E-1	1.59E0	-4.17E-2	-7.36E-1
3	1.59E0	1.08E-1	-6.67E-1	1.60E0	1.29E-1	-6.61E-1
1	1.14E0	4.98E-1	-4.31E-1	1.14E0	5.30E-1	-4.21E-1
6	6.89E-1	8.04E-1	-1.23E-1	6.93E-1	8.47E-1	-1.04E-1
23	2.37E-1	1.00E0	1.41E-1	2.41E-1	1.05E0	1.58E-1
36	1.86E-1	1.02E0	1.89E-1	1.90E-1	1.07E0	2.07E-1
54	6.61E-2	1.09E0	3.66E-1	7.04E-2	1.14E0	3.84E-1
171	-8.16E-3	1.14E0	4.57E-1	-8.17E-3	1.20E0	4.80E-1

100% PWR							
JT	DX	DY	DZ	JT	DX	DY	DZ
2	2.92E-3	-8.67E-1	1.68E-3	16	1.60E0	-3.92E-2	-7.38E-1
7	8.49E-2	-9.60E-1	1.72E-2	3	1.60E0	1.33E-1	-6.64E-1
5	2.56E-1	-9.69E-1	8.86E-2	1	1.15E0	5.39E-1	-4.25E-1
15	3.98E-1	-9.88E-1	5.29E-2	6	6.97E-1	8.61E-1	-1.15E-1
11	6.55E-1	-8.65E-1	-2.73E-1	23	2.46E-1	1.07E0	1.50E-1
12	9.19E-1	-7.01E-1	-5.99E-1	36	1.94E-1	1.09E0	1.99E-1
13	1.14E0	-5.07E-1	-9.26E-1	54	7.48E-2	1.17E0	3.76E-1
14	1.26E0	-3.64E-1	-9.54E-1	171	-8.17E-3	1.13E0	4.76E-1

ADDED: w. 201 7/15/82

CHKD: AL Ny 2 7/15/82

PREPARED BY RP Potelle DATE 6/25/82 DOC. NO. \_\_\_\_\_  
 REVIEWED BY ME Pincus DATE 6/25/82 PAGE NO. \_\_\_\_\_

*Kenly Kies*

Babcock & Wilcox

ATTCH # 4

POS. 71037.1 0.88

Nuclear Power Generation Division

GENERAL CALCULATIONS

DAVIS-BESSE SURGE PIPING SEISMIC DISPLACEMENTS

inches	X-DIR EQ			Y-DIR EQ		
JT	DX	DY	DZ	DX	DY	DZ
2	5.10E-2	1.99E-4	1.12E-2	3.07E-4	3.22E-3	6.01E-4
7	5.50E-2	2.64E-3	1.14E-2	7.95E-4	3.85E-3	9.42E-4
5	5.50E-2	3.53E-2	1.42E-2	7.96E-4	6.27E-3	2.98E-3
15	5.61E-2	5.39E-2	1.87E-2	2.02E-3	8.74E-3	4.52E-3
11	6.81E-2	6.03E-2	1.87E-2	9.01E-3	1.08E-2	4.52E-3
12	8.12E-2	5.96E-2	1.87E-2	1.19E-2	9.48E-3	4.51E-3
13	8.96E-2	5.13E-2	1.87E-2	9.70E-3	6.43E-3	4.49E-3
14	4.70E-2	4.84E-2	1.56E-2	7.15E-3	4.93E-3	4.05E-3
18	5.08E-2	4.84E-2	1.25E-2	3.93E-3	4.92E-3	3.28E-3
16	1.46E-1	4.82E-2	1.60E-2	2.11E-3	4.90E-3	4.73E-3
3	1.89E-1	1.12E-2	2.47E-2	2.44E-3	6.18E-3	7.80E-3
1	1.87E-1	7.68E-2	4.80E-2	2.45E-3	1.43E-2	1.70E-2
6	1.90E-1	5.90E-2	3.82E-2	2.45E-3	1.28E-2	2.06E-2
23	1.91E-1	1.42E-2	1.20E-2	2.44E-3	6.99E-3	2.25E-2
36	1.92E-1	1.20E-2	9.39E-3	2.21E-3	6.38E-3	2.26E-2
54	1.97E-1	9.13E-3	2.26E-3	1.71E-3	5.77E-3	2.25E-2
71	2.00E-1	5.52E-3	2.96E-3	1.89E-3	6.15E-3	2.25E-2

Z-DIR EQ							
JT	DX	DY	DZ	JT	DX	DY	DZ
2	1.31E-2	5.50E-4	6.34E-2	16	7.24E-3	1.55E-2	4.66E-2
7	1.50E-2	9.37E-3	6.25E-2	3	1.31E-2	8.36E-3	7.51E-2
5	1.50E-2	3.63E-2	5.81E-2	1	1.30E-2	3.67E-2	1.31E-1
15	2.23E-2	5.57E-2	5.87E-2	6	1.29E-2	5.53E-2	1.59E-1
11	4.99E-2	7.29E-2	5.87E-2	23	1.28E-2	6.06E-2	1.70E-1
12	6.14E-2	6.64E-2	5.86E-2	36	1.11E-2	6.02E-2	1.70E-1
13	5.66E-2	2.87E-2	5.85E-2	54	9.51E-3	5.98E-2	1.67E-1
14	4.21E-2	1.56E-2	4.28E-2	71	8.11E-3	6.19E-2	1.63E-1
18	1.79E-2	1.55E-2	2.04E-2				

ADDED: *Kenly Kies* 7/15/82  
 CHKD: *Alan* 7/15/82

PREPARED BY *MEP* DATE 7-1-82 DOC. NO. \_\_\_\_\_  
 REVIEWED BY *JT* DATE 7-1-82 PAGE NO. \_\_\_\_\_



Docket No. 50-346

License No. NPF-3

Serial No. 1-496

February 1, 1985



RICHARD P. CROUSE  
Vice President  
Nuclear  
14191 259-5221

Mr. James G. Keppler, Regional Administrator  
Region III  
United States Nuclear Regulatory Commission  
799 Roosevelt Road  
Glen Ellyn, Illinois 60137

Dear Mr. Keppler:

This letter transmits the information that Toledo Edison agreed to provide in the January 28, 1985 meeting at the Nuclear Regulatory Commission (NRC) Glen Ellyn offices, between members of the NRC Region III staff and Toledo Edison representatives, relative to a failed snubber found on the pressurizer surge line at the Davis-Besse Nuclear Power Station, Unit 1.

Information was requested to be submitted by February 1, 1985, to demonstrate the adequacy of the Nuclear Class 1 stanchion to pipe weld.

The requested information is provided in attached Calculation number M8, Revision 01, in its entirety.

These calculations show the stanchion to pipe weld to be acceptable for the design conditions.

Based on the above, Toledo Edison is confident of the adequacy of the new support design, which was completed during the refueling outage as corrective action relative to the failed snubber.

We trust that this fully addresses the concerns expressed by members of the NRC staff in the January 28, 1985 meeting, and thus removes the constraint to proceed to full power operation.

Very truly yours,

A handwritten signature in cursive script, appearing to read 'R. Crouse'.

RPC:JKW:lrh

cc: DB-1 NRC Resident Inspector

FEB 4 1985





# CALCULATION COVER SHEET

PROJECT DAVIS-BESSE, UNIT-1 JOB NO. 12501 CALC. NO. M-8  
 SUBJECT PRESSURIZER SURGE LINE - CLASS-I PIPING ANALYSIS

COMPUTER PROGRAM:  NONE  SCP  OTHER  
 PROGRAM NO(S) ME-101, ME-210 VERSION/RELEASE NO. K1/4-15-84, 5/12-15-82  
 ATT. NO. 1 = 3 SHTS., ATT. NO. 2 = 14 SHTS., ATT. NO. 3 = 4 SHTS., ATT. NO. 4 = 2 SHTS., ATT. NO. 5 = 1 SHT.

TOTAL NO. OF SHEETS 8  
(Excluding Attachments)

PRELIMINARY CALC.  COMMITTED PRELIMINARY CALC.  
 SUPERSEDED CALC.  FINAL CALC.

NO.	DESCRIPTION	BY	DATE	CHKD.	DATE	APPROVED	DATE
01	PRESSURIZER SURGE LINE	SPL	1-31-85	<del>SL</del>	1-31-85	MSW -	1-31-85

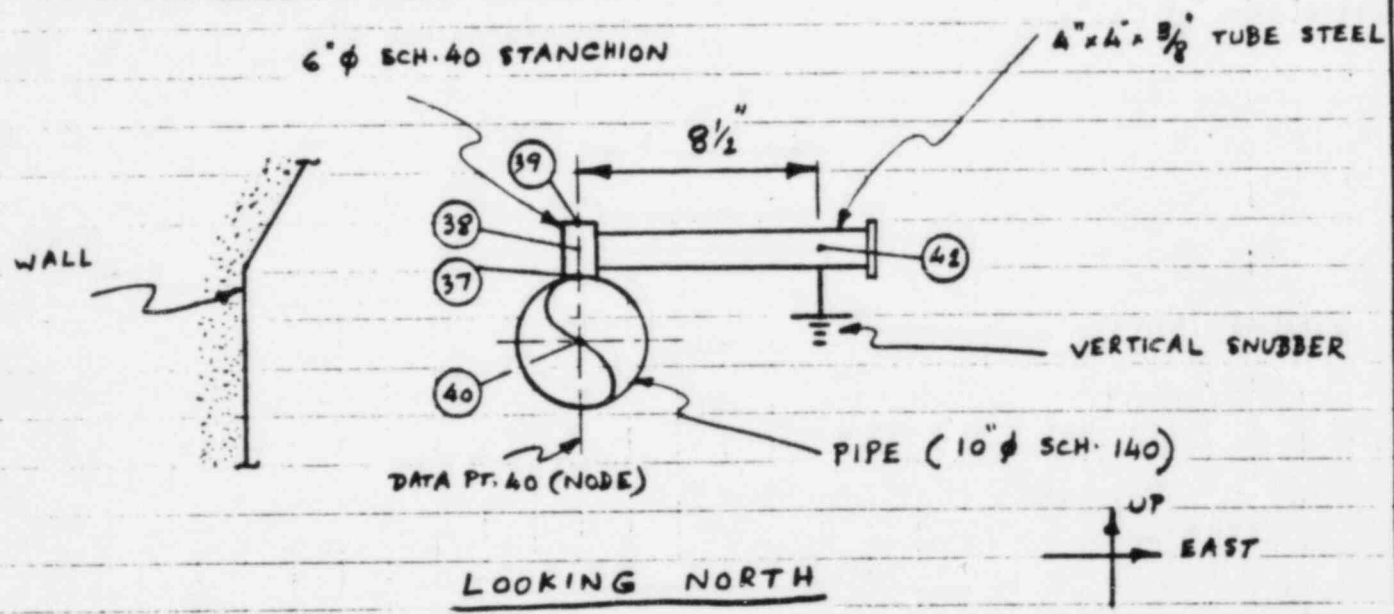
### REVISIONS



# CALCULATION SHEET

JOB NO. 12501 DAVIS-BESSE, UNIT. 1	CALC. NO. M-8	REV. NO. 01	SHEET NO. 1
PRESSURIZER SURGE LINE		ORIGINATOR	DATE
		DATE	CHECKED
siva p. lingam		1-30-85	1-31-85

AS-BUILT CONDITION OF VERTICAL SNUBBER (PSU-RI) ON PRESSURIZER SURGE LINE



THE PURPOSE OF THIS CALCULATION IS TO CHECK THE PRIMARY PIPE STRESSES (CATASTROPHIC FAILURE) ON THE PRESSURIZER SURGE LINE FOR LONG-TERM OPERATION. CRITERIA EXCLUSIVE OF FINAL CLASS-I FATIGUE ANALYSIS.

ORIGINAL ANALYSIS OF THIS LINE WAS DONE BY B&W ON MAY 3, 1974 (STRESS ANALYSIS OF SURGE LINE, REPORT #6, REV. 1). ALL POINTS WERE ANALYZED BY USING THE SIMPLIFIED METHOD PER USAS B31.7, 1968 (1-705 & TABLE D-201, APPENDIX-D). IN THIS ANALYSIS, VERTICAL SNUBBER PSU-RI IS CONSIDERED AT THE CENTER LINE OF PIPE.

AS PER NCR 84-180, THE WEST SIDE VERTICAL SNUBBER (CLOSE TO THE WALL) FAILED. THE REASON OF FAILURE IS DETERMINED AS THERMAL INTERFERENCE BETWEEN WALL AND THE FAILED SNUBBER. AS A RESULT, BECHTEL MODIFIED THIS SNUBBER AS SHOWN ABOVE TO MAKE THE REMAINING SNUBBER ALONE CAPABLE OF CARRYING THE ORIGINAL LOAD. HOWEVER THIS MODIFICATION CAUSES AN UNBALANCED MOMENT (TORSION ON PIPE) AT PIPE AND STANCHION INTERFACE.

GPO 2706 REV. 4/82 (ED-89)



# CALCULATION SHEET

JOB NO. 12501 DAVIS-BESSE, UNIT 1 CALC. NO. M-8		REV. NO. 01	SHEET NO. 2
PRESSURIZER SURGE LINE			
ORIGINATOR	DATE	CHECKED	DATE
Siva P. Kingam	1-30-85	[Signature]	1-31-85

BECHTEL MADE TWO SEISMIC RUNS OF PRESSURIZER SURGE LINE (BY DECOUPLING FROM HOT LEG), ONE WITH VERTICAL SNUBBER AT THE CENTER LINE OF PIPE AND OTHER WITH VERTICAL SNUBBER MODELED THROUGH STANCHION & TUBE STEEL TO REFLECT THE MODIFICATION AND AT THE SAME TIME CONSIDER THE EFFECT OF AN UNBALANCED MOMENT ON THE SYSTEM AS A WHOLE. THE DYNAMIC PROPERTIES (EIGEN VALUES & EIGEN VECTORS) WERE THEN COMPARED FOR THE TWO RUNS. THERE IS NO SIGNIFICANT CHANGE IN THE DYNAMIC PROPERTIES LEADING TO THE CONCLUSION THAT THE RESULTS OF THE ORIGINAL ANALYSIS REMAIN UNCHANGED INCLUDING NOZZLE LOADS. THE COMPARISON OF EIGEN VALUES (NATURAL FREQUENCIES UP TO 33 Hz) IS SHOWN BELOW (SEE ATTACHMENT NO. 1):

WITH SNUBBER AT CENTER LINE OF PIPE	WITH SNUBBER ACTUALLY MODELED	DIFFERENCE IN %	SOURCE
10.031 Hz.	9.805 Hz.	2.253%	BECHTEL
16.265	16.327	0.381	COMPUTER
19.021	18.914	0.563	RUNS
21.418	20.054	6.37 (HIGHER MODE)	G7399(11-28-84) AND
27.479	26.913	2.06	G7397(11-28-84)

HOWEVER, THE EFFECT OF ADDITIONAL TORSION MOMENT HAS TO BE CHECKED LOCALLY BY ADDING TO THE ORIGINAL RESULTS. SINCE THE MAXIMUM SEISMIC STRESS AND MAX. PRIMARY STRESS INTENSITY (EQ. 9 OF 1-705.1 OF 831.7, 196P) OCCUR VERY CLOSE TO THIS SNUBBER (4 1/4" AWAY) IN THE ORIGINAL ANALYSIS, - THE CORRESPONDING MOMENTS ARE CONSIDERED AT THE SNUBBER AND ARE NOTED BELOW (B&W DATA PT. 3 - SEE ATTACHMENT #2):

LOAD TYPE	MOMENT IN INCH-POUNDS			TORSIONAL MOMENT (M <sub>x</sub> )
	M <sub>x</sub>	M <sub>y</sub>	M <sub>z</sub>	
WEIGHT	12615.24	4661.04	51039.12	-
SEISMIC (OBE)	8872.8	22011.6	244343.04	20400 (2400 x 8.5)
SEISMIC (SSE)	15204.6	38394.	451370.28	33150 (3900 x 8.5)
WEIGHT + OBE	21488.04	26672.64	295382.16	20400
WEIGHT + SSE	27819.84	43055.04	502409.4	33150



# CALCULATION SHEET

JOB NO. 12501 DAVIS-BESSE, UNIT-1 CALC. NO. M-8		REV. NO. 01	SHEET NO. 3
PRESSURIZER SURGE LINE			
ORIGINATOR Ira P. Dungan	DATE 1-30-85	CHECKED [Signature]	DATE 1-31-85

## EQUATION-9

$$B_1 \frac{PD_0}{2t} + B_2 \frac{D_0}{2I} M_x \leq 1.5 S_m \quad (\text{UPSET})$$

$$B_1 \frac{PD_0}{2t} + B_2 \frac{D_0}{2I} M_x \leq 2.25 S_m \quad (\text{FAULTED PER B\&W ANALYSIS})$$

SEE ATTACH. NO. 2, SHT. NO. 13

$B_1 = 1.0$  (elbow) (conservative as stanchion in run straight pipe) → Ref. Table D-201 of APP. D  
 NO hole in the run pipe - the cases only longitudinal pressure stress.

$P = 2500 \text{ psi}$

$D_0 = 10.75''$

$t = 1.0''$

$I = 368 \text{ in}^4$

$$M_x = \sqrt{(21488.04 + 20400)^2 + (26672.64)^2 + (295382.16)^2}$$

$= 299527.4 \text{ in-lb. (upset)}$

$$B_2 = 0.75 C_2 \cdot 0.75 \left[ \left( \frac{r}{r_1} \right)^{4/3} \right] M_x = \sqrt{(27819.84 + 33150)^2 + (43055.04)^2 + (502409.4)^2}$$

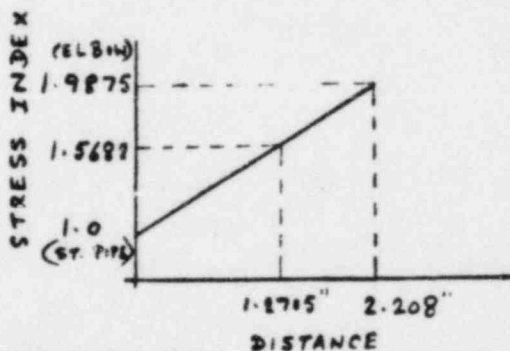
$= 507923.5 \text{ in-lb. (Faulted)}$

$B_2 = 1.9875$  (FOR ELBOW)

$B_2 = 1.5$  (FOR WTEE)

$$\sqrt{kt} = \sqrt{4.875 \times 1.0} = 2.208''$$

ACTUAL DISTANCE BETWEEN ELBOW  
 WELD & NEAREST STANCHION WELD  
 $= 0.9875''$



ASSUMING LINEAR DISTRIBUTION OF STRESS INDEX,

$$\frac{0.9875}{B} = \frac{2.208}{1.2705}$$

$B = 0.5682$

$B = 1.0 + 0.5682 = 1.5682$

$B_2 = 1.5682 \times 1.5 = 2.3523$  (multiplication of overlapping stress index is not required but is implied in note 2 of table D-201)

If assumed as an unreinforced tee (UTEE),  $i = \frac{0.9}{k^{2/3}}$  where

$k = \frac{kt}{r}$   
 (from class-2)

$k_n = 1.0, r = 4.875$

$i = 2.5877$

$B_1 = 2.5877$  (conservative - no hole in the run pipe)





# CALCULATION SHEET

JOB NO. 12501 DAVIS-BESSE, UNIT-1		CALC. NO. M-8		REV. NO. 01	SHEET NO. 4
PRESSURIZER SURGE LINE					
ORIGINATOR	DATE	CHECKED	DATE		
<i>Eva P. Lingam</i>	1-30-85	<i>Randall's</i>	1-31-85		

1  
2 CONSIDER THE HIGHEST VALUE OF  $B_2 = 2.5877$  (UTEE)

3  
4  $B_1 \frac{PD_1}{2L} + B_2 \frac{PD_2}{2L} M_2 \leq 1.5 S_m$

5  
6  
7  $1.0 \times \frac{2500 \times 10.75}{2 \times 1.0} + 2.5877 \times \frac{10.75}{2 \times 368} \times 299527.4 \leq 24,800 \text{ psi}$

8  
9  
10  $24,759 \leq 24,800 \text{ psi}$

11 (UPSET CONDITION)

12  
13  
14  
15  $1.0 \times \frac{2500 \times 10.75}{2 \times 1.0} + 2.5877 \times \frac{10.75}{2 \times 368} \times 507923.5 \leq 37,575 \text{ psi}$

16  
17  
18  $32,635 \leq 37,575 \text{ psi}$

19  
20 (Faulted condition)  
21 SEE ATTACH. NO. 2  
22  
23  
24  
25  
26  
27  
28  
29  
30  
31  
32  
33  
34  
35  
36



# CALCULATION SHEET

JOB NO. 12501 DAVIS-BESSE, UNIT-1 CALC. NO. M-8		REV. NO. 01	SHEET NO. 5
PRESSURIZER SURGE LINE			
ORIGINATOR	DATE	CHECKED	DATE
Siva P. Aringam	1-30-85	[Signature]	1-31-85

1 BECHTEL PROGRAM ME-210 ALSO CALCULATES THE LOCAL STRESSES  
 2 AT THE STANCHION & PIPE INTERFACE. Ref.: WRC-107

3  
 4  
 5 CHECKS 1A, 1B, 2A & 2B OF ME-210 DEAL WITH UPSET PRIMARY  
 6 (CIRCUMFERENTIAL), FAULTED PRIMARY (CIRCUMFERENTIAL), UPSET  
 7 PRIMARY (LONGITUDINAL) & FAULTED PRIMARY (LONGITUDINAL)  
 8 RESPECTIVELY. (SEE SHT. 4 OF ATTACH. NO. 3)

9  
 10 1A →  $\frac{P D_0}{2t} + \text{stresses from ME-210} = 14,030 \text{ psi}$

11  
 12  $13437.5 + \text{stresses from ME-210} = 14,030$

13  
 14  $\text{stress from ME-210} = 14030 - 13437.5 = 592.5$

15 Applying stress index of 2.5877,

16  
 17  
 18  
 19 1A →  $13437.5 + (2.5877 \times 592.5) = 14,971 \text{ psi} (< 24,720 \text{ psi})$

20  
 21 1B →  $13437.5 + 2.5877 \times (14400 - 13437.5) = 15,929 \text{ psi}$   
 22  $(< 37,575 \text{ psi})$   
 23  $B6W = 2.255$

24 2A →  $\frac{P D_0}{4t} + \text{stresses from ME-210 including ME-101 stresses} = 12,970$

25  
 26  $6718.75 + \text{stresses from ME-210 including ME-101 stresses} = 12,970$

27  
 28  $\text{stresses from ME-210 including ME-101 stresses} = 6251.25$

29 Applying stress index of 2.5877,

30  
 31  
 32 2A →  $6718.75 + (2.5877 \times 6251.25) = 22,896 \text{ psi} (< 24,720 \text{ psi})$

33  
 34 2B →  $6718.75 + 2.5877 \times (17390 - 6718.75) = 34,332 \text{ psi}$   
 35  $(< 77,575 \text{ psi})$   
 36



# CALCULATION SHEET

JOB NO. 12501	CALC. NO. M-8	REV. NO. 01	SHEET NO. 6
DAVIS-BESSE UNIT ONE		DATE 1-31-85	CHECKED Siva P. dungam
ORIGINATOR J. J. P. [Signature]	DATE 1-31-85	CHECKED Siva P. dungam	DATE 1-31-85

REANALYSIS USING CODE CASE N-411 RESULTED IN THE SYSTEM EXPERIENCING THE FOLLOWING REDUCTIONS IN DISPLACEMENTS, LOADINGS AND STRESSES (WHEN COMPARED TO THE ORIGINAL ANALYSIS):  
 (CODE CASE N-411 CALLS OUT FOR 5% DAMPING FOR 0-10cps, A LINEAR DECREASE FROM 5% TO 2% DAMPING FOR 10-20cps, AND 2% DAMPING FOR FREQUENCIES GREATER THAN 20cps - SEE ATTACHMENT #4)

SUPPORTS: @ BECHTEL D.P. (42) : 64% REDUCTION IN LOAD (Y-SNUBBER) - PSURI  
 @ " " (55) : 11% REDUCTION IN LOAD (X-SNUBBER)  
 @ " " (60) : 70% REDUCTION IN LOAD (Z-SNUBBER)

DISPLACEMENTS: ALL DATA POINTS: 60% TO 75% REDUCTION IN DISPLACEMENT

MOMENTS IN THE PIPE: FOR M<sub>X</sub>: 51% TO 70% REDUCTION IN LOADING  
 FOR M<sub>Y</sub>: 62% TO 70% REDUCTION IN LOADING  
 FOR M<sub>Z</sub>: 48% TO 70% REDUCTION IN LOADING

STRESSES IN THE PIPE: AN OVERALL REDUCTION OF 64% TO 69% IN STRESSES.

NOZZLE LOADS: FOR HOT LEG (BECHTEL D.P. (5)): 49% TO 69% REDUCTION IN LOADING  
 FOR PRESSURIZER (BECHTEL D.P. (100)): 11% TO 70% REDUCTION IN LOADING

THE REDUCTIONS SHOWN ABOVE REFLECT OBE AND SSE CONDITIONS. IN GENERAL, THE MORE SIGNIFICANT LOADINGS EXPERIENCED APPROXIMATELY A 65% REDUCTION IN LOADINGS.

REFER TO: SNUMB N° 60087 DATED 1-30-85 - OBE USING CODE CASE N-411  
 SNUMB N° 60096 DATED 1-31-85 - SSE USING CODE CASE N-411



# CALCULATION SHEET

JOB NO. 12501 DAVIS-BESSE, UNIT-1 CALC. NO.

M-8

REV. NO.

01

SHEET NO.

7

PRESSURIZER SURGE LINE

ORIGINATOR

DATE

CHECKED

DATE

Siva P. Lingam

1-31-85

R. J. [Signature]

1-31-85

IT IS EVIDENT FROM THE PREVIOUS SHEET THAT AT LEAST 40% REDUCTION IN Y-LOAD, MOMENTS & STRESSES CAN BE ATTAINED FOR CODE CASE N-411 WHICH PERMITS THE USE OF HIGHER DAMPING (5% TO 2% OF CRITICAL DAMPING). APPLYING THIS REDUCTION IN EQUATION-9, WE GET THE FOLLOWING:

$$B_1 \frac{PD_0}{2t} + B_2 \frac{D_0}{2I} M_i \leq 1.5 S_m$$

$$1.0 \times \frac{2500 \times 10.75}{2 \times 1.0} + 2.5877 \times \frac{10.75}{2 \times 368} \cdot (299527.4) \cdot 0.6 \leq 24,800 \text{ psi}$$

$$20,231 \leq 24,800 \text{ psi}$$

(upset condition)

$$1.0 \times \frac{2500 \times 10.75}{2 \times 1.0} + 2.5877 \times \frac{10.75}{2 \times 368} \cdot (507923.5) \cdot 0.6 \leq 37,575 \text{ psi}$$

$$24,956 \leq 37,575 \text{ psi}$$

(faulted condition)

IN A SIMILAR WAY, ME-210 STRESSES (CHECKS 1A, 1B, 2A & 2B) WILL BE MUCH WITHIN THE ALLOWABLES.

## CONCLUSIONS

EVEN WITH CONSERVATIVE PRIMARY STRESS INDICES ( $B_1$  FOR AN ELBOW ALTHOUGH STANCHION IS ON A STRAIGHT PIPE AND  $B_2$  FOR AN UNREINFORCED TEE ALTHOUGH THERE IS NO HOLE IN THE PIPE), CONDITION FOR CATASTROPHIC FAILURE IS SATISFIED FOR LONG TERM OPERATION. IN ADDITION, WE HAVE SHOWN THAT SEISMIC ANALYSIS USING 5% DAMPING WOULD REDUCE STRESS LEVELS AN ADDITIONAL 18% (AT LEAST) AND MATERIAL TEST REPORTS SHOW THE ACTUAL YIELD STRENGTH TO BE GREATER THAN THE ALLOWABLE DESIGN VALUES.



TITLE 1 PRESSUR, SURGE LINE  
 PROJECT NUMBER 1 12581  
 PROBLEM NUMBER 1 40  
 USER 1 SPL  
 LOAD CASE 1 BE181

EIGEN SOLVER 1 DETERMINANT SEARCH

FREQUENCIES FOR THE BE181 LOAD CASE (CP8)  
 19.0315238 16.2646013 19.0205380 21.4163087 27.4791067

EIGEN VALUES (natural frequencies)

PERIODS FOR THE BE181 LOAD CASE (SEC)  
 .0996877 .0616629 .0525748 .0866890 .0363913

SEISMIC

WITHOUT BROKEN SNUBBER

PRESSURIZER SURGE LINE

ATTACHMENT NO. 1

PROB. NO. M-8 (01)

SHT. NO. 1 OF 3

Added by: Steve P. Dingman Date: 1-30-85

CHK'd by: David J. [Signature] Date: 1-31-85

FREQUENCIES AND PERIODS

TITLE 1 PRESSUR. SURGE LINE  
 PROJECT NUMBER 1 12501  
 PROBLEM NUMBER 1 M8  
 USER 1 SPL  
 LOAD CASE 1 BEIS1

EIGEN SOLVER 1 DETERMINANT SEARCH

FREQUENCIES FOR THE BEIS1 LOAD CASE (CP8) EIGEN VALUES (natural frequencies)

9.8041666	16.3247980	16.9136366	20.0543249	26.9129386
-----------	------------	------------	------------	------------

SEISMIC

PERIODS FOR THE BEIS1 LOAD CASE (SEC)

.1019975	.0612490	.0528719	.0498686	.0371568
----------	----------	----------	----------	----------

WITH BROKEN SNUBBER

2.3% - first frequency

PRESSURIZER SURGE LINE

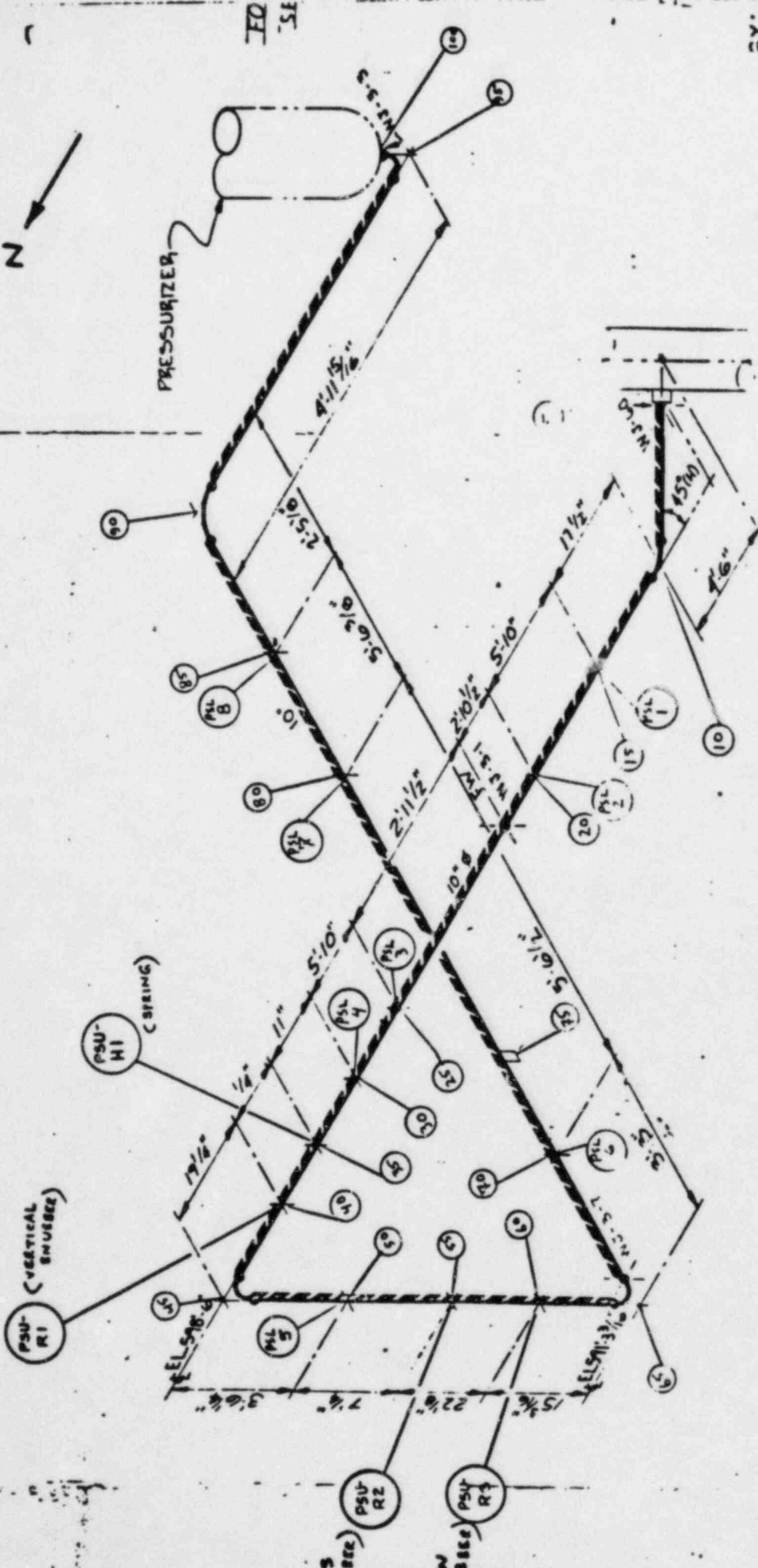
ATTACHMENT NO. 1

PROB. NO. M-8(01)

SHT. NO. 2 OF 3

Added by: Siva P. Arangan Date: 1-30-85

chkd by: [Signature] Date: 1-31-85



BY: \_\_\_\_\_  
 CHKD \_\_\_\_\_  
 APPR \_\_\_\_\_

PROB. NO. M-8 (01)

ATTACHMENT # 1

SHT. NO. 3 OF 3

Added by: Sura J. Singam Date: 1-30-85

Checked by: [Signature] Date: 1-31-85

Randy Kies A.S.N: SIVA LINGAM

**Babcock & Wilcox**

a WILCOX COMPANY

PG-2027-1 9-82

Nuclear Power Generation Division

**GENERAL CALCULATIONS**

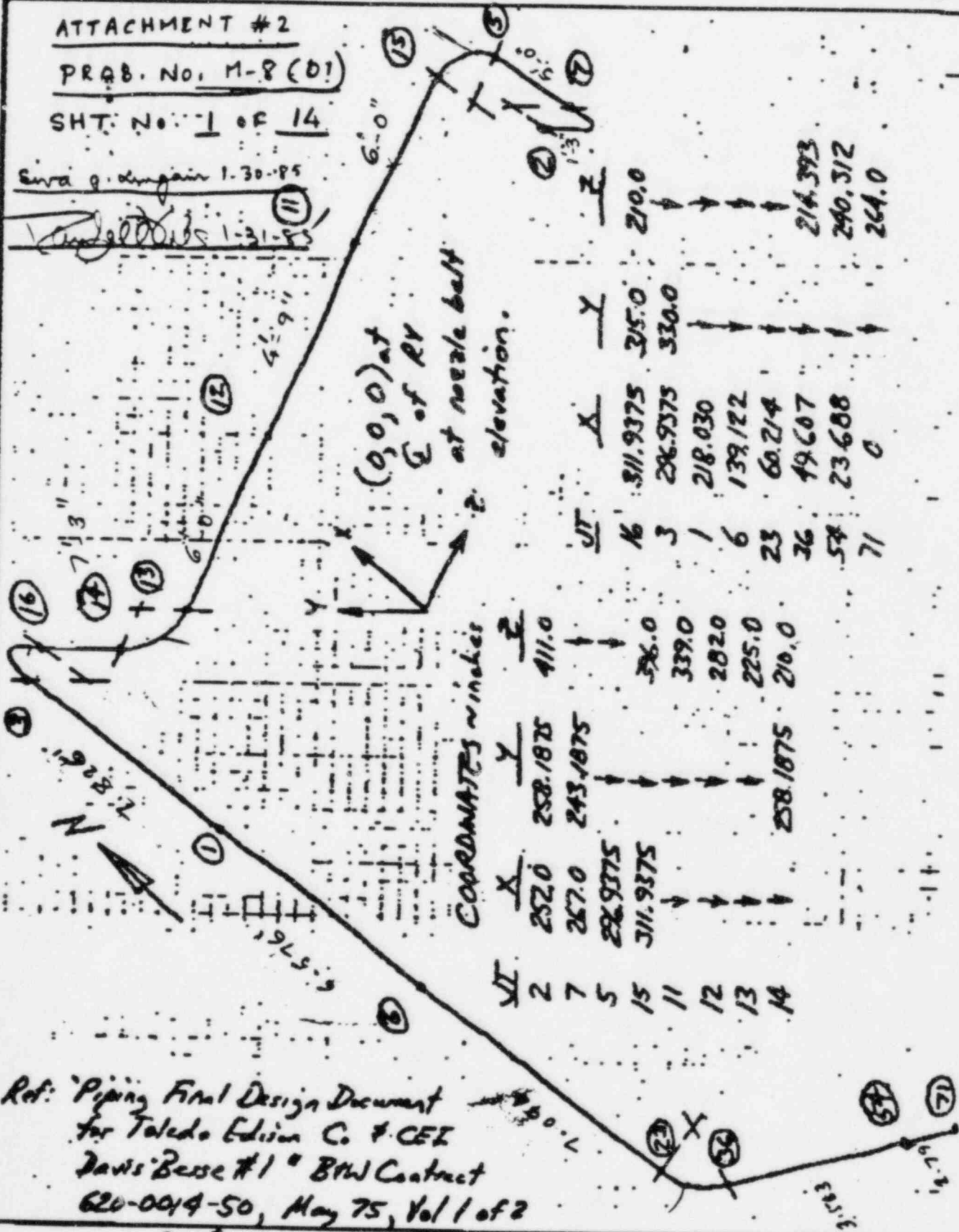
ATTACHMENT #2

PRAB. NO. M-8 (D1)

SHT. NO. 1 OF 14

Added by: Siva Lingam 1-30-85

Checked by: [Signature] 1-31-85



COORDINATES inches

ST.	X	Y	Z
2	252.0	258.1875	411.0
7	267.0	243.1875	
5	286.9375		386.0
15	311.9375		339.0
11			282.0
12			225.0
13			210.0
14		258.1875	
16	311.9375	335.0	210.0
3	286.9375	330.0	
1	218.030		
6	139.122		
23	60.214		
26	49.607		
54	23.688		
71	0		

Ref: Piping Final Design Document  
for Toledo Edison Co. & CEI  
Davis Base #1 - BTW Contract  
620-0014-50, May 75, Vol 1 of 2

PREPARED BY RP Petukhin DATE 6/25/82 DOC. NO. \_\_\_\_\_  
 REVIEWED BY \_\_\_\_\_ DATE \_\_\_\_\_ PAGE NO. \_\_\_\_\_



SURGE LINE SUPPRESSOR/HANGER LOADS

(See Figure 4)

<u>Type of Loading</u>	<u>Forces (Kips)</u>			<u>Moments (Ft-Kips)</u>		
	Fx	Fy	Fz	Mx	My	Mz
Dead Weight	0	-3.8	0	0	0	0
OBE	7.5	2.4	3.2	0	0	0
DSE	13.7	3.9	5.8	0	0	0

PROB. NO. M-8(01)

ATTACHMENT # 2

SHT. NO. 2 OF 14

Added by: Siva P. Lingam Date: 1-30-85

checked by: Randall Date: 1-31-85

## Introduction

This report contains the stress analysis of the surge line. All points in the surge line are analyzed including the bimetallic weld at the hot leg surge nozzle. The analysis of the intersection at the pressurizer end is contained in pressurizer Design Report "Surge Nozzle Analysis". The branch intersection analysis is contained in this Design Report under a different section "Surge Nozzle".

## Results

All points in the surge line satisfy primary stress limits either in accordance with equation 9 or per Appendix F of the B31.7 code.

### Per equation 9 simplified analysis

Maximum Primary Stress = 22074.96 psi  $< 1.5 S_m = 24800$ . psi @ 670°F  
Maximum for an elbow (joint 3) = 22074.96 psi  $< 24800$ . psi  
Maximum for a straight (joint 1) = 10220.54 psi  $< 24800$ . psi

All points in the surge line do not satisfy the  $3 S_m$  primary plus secondary stress limit. However, calculations are provided showing that an Elastic-Plastic Analysis is applicable.

Maximum primary plus secondary stress (Bimetallic Weld) (joint 54)  
= 75997.62 psi  $> 3 S_m = 50100$ . psi @ 650°F (PAGE C-11)  
Maximum primary plus secondary stress for an elbow (joint 2)  
= 67368.71 psi  $> 50100$ . psi (PAGE C-13)  
Primary plus secondary stress for cycles occurring more than 240 times:  
Joint 2 = 36860.4 psi  $< 3 S_m = 50100$ . psi  
Joint 3 = 33787.9 psi  $< 3 S_m = 50100$ . psi  
Joint 19 = 36929.7 psi  $< 3 S_m = 50100$ . psi

### Maximum Usage Factors:

Usage Factor at Bimetallic weld (joint 2) = 0.87  $< 1.0$  = allowable  
Usage Factor for joint 3 = 0.675  $< 1.0$  = allowable  
Usage Factor for joint 19 = 0.736  $< 1.0$  = allowable

PROB. NO. M-8 (01)

ATTACHMENT # 2

SHT. NO. 3 OF 14

Added by: Siva P. Lingam Date: 1-30-85

Checked by: [Signature] Date: 1-31-85

Ref: "Stress Analysis of Surge Line, Report #6, for Toledo Edison Company, Davis-Besse, "620-0014-50, Jun 1972 (Rev 1, 5/3/74). RP 7/2/82"

This section demonstrates that all points in the surge line satisfy primary stress limits according to either Equation 9 or Appendix F of B31.7. A flexibility analysis was run for dead weight, and either (x + y) or (z + y) earthquakes. The moments generated by this analysis are shown on Pages A-4 thru A-7.

These moments are then combined with design pressure of 2500 psi to generate primary stresses in accordance with equation 9 using applicable indices. The indices used are listed in the stress output. Two cases were ran: 1) pressure, deadload, (x + y) earthquake; 2) pressure, deadload, (z + y) earthquake.

The results are then compared to  $1.5 S_m$  at  $670^{\circ}\text{F}$  for SA-376-TP-316 (straights) and A-403-WP-316 (bends). Part of the surge line actually has a design temperature of  $650^{\circ}\text{F}$ .  $S_m = 16,500$  psi. for both materials thus the allowable primary stress is  $1.5 (16,500.) = 24,800.$  psi.

The largest primary stress is 22074.96 psi at joint 3 for the x + y earthquake at an elbow juncture. Complete results are tabulated on Page A-8 and 9.

PROB. NO. M-8 (01)

ATTACHMENT # 2

SHT. NO. 4 OF 14

Added by: Siva S. Singam date: 1-30-85

ch'kd by: Vandana date: 1-31-85

STRESS INDICES

(JOINT CLASSIFICATION)

TYPE

DESCRIPTION

- 1 STRAIGHT PIPE, REMOTE FROM WELDS OR OTHER DISCONTINUITIES
- 2 GIRTH BUTT WELD BETWEEN STRAIGHT PIPE OR BETWEEN PIPE AND BUTT-WELDING COMPONENTS. FLUSH
- 3 GIRTH BUTT WELD BETWEEN STRAIGHT PIPE OR BETWEEN PIPE AND BUTT-WELDING COMPONENTS. AS WELDED
- 4 GIRTH FILLET WELD TO SOCKET WELD FITTINGS, SLIP-ON FLANGES, OR SOCKET-WELDING FLANGES
- 5 LONGITUDINAL BUTT WELDS IN STRAIGHT PIPE. FLUSH
- 6 LONGITUDINAL BUTT WELDS IN STRAIGHT PIPE. AS WELDED
- 7 TAPERED TRANSITION JOINTS PER SUBPAR. 1-727.4.2(C) AND FIG.1-727.3.1
- 8 BRANCH CONNECTIONS PER SUBDIV. 1-704.3
- 9 CURVED PIPE OR BUTT-WELDING ELBOWS PER USAS B16.9, USAS B16.28, OR MSS SP48
- 10 BUTT-WELDING TEES PER USAS B16.9 OR MSS SP48
- 11 BUTT-WELDING REDUCERS PER USAS B16.9 OR MSS SP48

PROB. NO. M-8(01)

ATTACHMENT # 2

SHT. NO. 5 OF 14

Added by: Siva P. Kingam date: 1-30-85

ch'kd by: [Signature] date: 1-31-85



TOLEDO SURGE LINE

SECTION PROPERTIES

SECTION	OUTSIDE RADIUS (IN)	OUTSIDE DIAMETER (IN)	THICKNESS (IN)	MOMENT OF INERTIA (IN <sup>4</sup> )	BEND RADIUS (IN)	DESCRIPTION
1	5.3756	10.7500	1.0000	367.805		SURGE LINE ST.
2	5.3750	10.7500	1.0000	367.805	15.0000	SURGE LINE CR.

PROB. NO. M-8 (01)

ATTACHMENT # 2

SHT. NO. 6 OF 14

Added by: Siva P. Lingam date: 1-30-85

ch'ed by: Randall's date: 1-31-85

## DEAD LOAD MOM. (FT-LBS)

JOINT	M(1)	M(2)	M(3)
1	1051.27	190.21	2956.52
2	48.76	575.73	5475.20
3	1051.27	388.42	4253.26
5	11.08	462.85	162.55
6	1051.27	8.18	4545.22
7	11.08	538.05	3365.43
11	4089.10	120.24	1611.11
12	3647.55	121.16	1611.11
13	352.88	362.57	1611.11
14	870.89	426.09	1674.64
15	1437.53	361.65	1611.11
16	1013.59	426.09	1915.25
18	942.24	426.09	1794.94
23	1051.27	206.20	512.79
36	780.86	251.45	142.81
54	1677.71	426.33	2599.77

PROB. NO. M-8 (01)

ATTACHMENT NO. 2

SHT. NO. 7 OF 14

Added by: Siva P. Singam Date: 1-30-85

ch'ed by: Rangababu's Date: 1-31-85

JOINT	M(X)	M(Y)	M(Z)
1	552.91	5774.54	13644.50
2	1091.14	3150.34	8890.93
3	552.91	1462.31	19615.80
5	803.58	3091.78	5837.96
6	552.91	2393.62	4150.89
7	803.58	2984.70	7184.48
11	2332.90	3832.01	5367.42
12	2374.86	4664.09	5367.42
13	1686.72	2087.29	5367.42
14	1926.39	1927.74	5317.74
15	316.78	2491.91	5367.42
16	762.69	1927.74	10767.60
18	2770.96	1927.74	5636.34
23	552.91	2913.27	5693.95
36	737.77	2680.95	7000.95
54	4852.93	7976.70	10344.10

ATTACHMENT # 2

PROB. NO. M-8(01)

SHT. NO. 8 OF 14

Added by: Siva P. Lingam Date: 1-30-85

Checked by: Randall's Date: 1-31-85

Y EARTHQUAKE MOMENTS (FT-LBS)

JOINT	MIXI	MIIYI	MIZI
1	186.49	1510.79	1028.76
2	314.37	812.57	1047.93
3	186.49	371.99	746.12
5	403.67	702.55	235.87
6	186.49	613.02	711.74
7	403.67	744.95	767.22
11	1145.29	1185.50	496.78
12	881.24	1481.76	496.78
13	948.49	471.82	496.78
14	1066.14	571.52	301.27
15	117.58	410.45	496.78
16	221.74	571.52	445.15
18	773.38	571.52	711.91
23	186.49	780.29	522.38
36	241.32	811.28	768.08
54	1042.90	1347.92	1561.55

PROB. NO. M-8(01)

ATTACHMENT # 2

SHT. NO. 9 OF 14

Added by: Siva P. Dingam date: 1-30-85

ch'kd by: Parthasarathy's date: 1-31-85

A-6



JOINT	H1X1	H1Y1	H1Z1
1	2433.66	7961.39	4184.12
2	3388.12	8215.63	10092.18
3	2433.64	2335.07	3456.56
5	3399.73	2767.35	1725.99
6	2433.64	4134.19	2241.79
7	3399.79	5695.74	6636.94
11	6557.46	5244.52	943.78
12	8413.33	5365.99	943.78
13	6341.99	2892.85	943.78
14	4395.61	3892.77	1690.56
15	1445.88	3424.23	943.78
16	1675.18	3892.77	2984.58
18	3877.61	3892.77	4513.19
23	2433.64	3949.75	1820.32
36	2629.71	4271.87	1651.72
54	4643.93	9844.89	3448.78

PROB. NO. M-8 (01)

ATTACHMENT # 2

SHT. NO. 10 OF 14

Added by: Siva P. Lingam Date: 1-30-85

ch'kd by: Randall's Date: 1-31-85

EQUATION 9 - PRIMARY STRESSES

CONDITION . . . PRESSURE, X+Y EQ , DEAD WT.  
EQ 9

MOMENTS IN FT-LBS.

STRESSES IN PSI.

JOINT	TYPE	HX	HY	HZ	H(II)	B1	B2	PRIMARY STRESS	RATIO TO ALLOWABLE
1	1	1790.67	7475.54	18429.78	19968.65	.5	1.0	10220.54	.413
2	9	1454.47	4538.64	15454.06	16172.28	1.0	2.0	19074.58	.771
3	9	1790.67	2222.71	24615.18	24780.11	1.0	2.0	22874.96	.892
5	9	1223.34	4256.18	6236.38	7648.80	1.0	2.0	16103.60	.651
6	1	1790.67	3011.64	9487.85	10039.13	.5	1.0	8479.26	.343
7	9	1223.34	4271.70	11317.13	12158.18	1.0	2.0	17675.41	.714
11	1	7487.29	5137.75	7475.31	11761.64	.5	1.0	8781.32	.355
12	1	5902.85	6267.01	7475.31	11950.11	.5	1.0	8814.37	.356
13	9	2988.09	2921.68	7475.31	8564.18	1.0	2.0	16422.67	.664
14	9	3863.42	2925.35	7293.65	8756.77	1.0	2.0	16489.88	.666
15	9	1871.89	3264.00	7475.31	8368.87	1.0	2.0	16354.59	.661
16	9	1998.82	2925.35	13127.99	13597.57	1.0	2.0	18177.13	.734
18	1	4486.58	2925.35	8143.19	9746.68	.5	1.0	8427.97	.341
23	9	1790.67	3639.76	6729.12	7980.96	1.0	2.0	16219.38	.655
36	9	1759.15	3743.67	7911.84	8927.17	1.0	2.0	16549.19	.669
54	3	6773.54	9750.94	14505.42	18744.84	.5	1.0	10085.93	.404

PROB. NO. M-8(01)

ATTACHMENT # 2

SHT. NO. 11 OF 14

Added by: Siva P. Singam date: 1-30-85

ch'ed by: Raymond Lee's date: 1-31-85

TOLEDO SURGE LINE

EQUATION 9 - PRIMARY STRESSES

CONDITION . . . PRESSURE , Z+Y EQ , DEAD WT.  
EQ 9  
MOMENTS IN FT-LBS.

STRESSES IN PSI.

JOINT	TYPE	MOMENTS IN FT-LBS.				STRESSES IN PSI.		PRIMARY STRESS	RATIO TO ALLOWABLE
		MX	MY	MZ	M(II)	B1	B2		
1	1	3671.40	9662.39	8969.40	13685.44	.5	1.0	9110.69	.368
2	9	3751.45	9603.93	16655.23	19588.40	1.0	2.0	20265.32	.819
3	9	3671.40	3095.47	8455.93	9724.40	1.0	2.0	16827.08	.680
5	9	3819.54	3932.76	2124.41	5879.51	1.0	2.0	15486.89	.626
6	1	3671.40	4752.21	7498.75	9606.98	.5	1.0	8403.47	.340
7	9	3819.54	6978.74	10769.59	13389.39	1.0	2.0	18104.56	.731
11	1	11712.35	6550.26	3051.67	13762.19	.5	1.0	9132.15	.369
12	1	12942.72	6968.91	3051.67	15013.08	.5	1.0	9351.51	.378
13	9	7643.36	3727.24	3051.67	9034.71	1.0	2.0	16586.68	.670
14	9	6332.64	4890.38	3656.47	8797.04	1.0	2.0	16503.83	.667
15	9	3000.99	4196.32	3051.67	5993.98	1.0	2.0	15526.79	.627
16	9	2910.51	4890.38	5344.89	7807.36	1.0	2.0	16158.87	.653
18	1	5593.15	4890.38	7020.04	10221.55	.5	1.0	8511.25	.344
23	9	3671.40	4936.24	2855.49	6782.38	1.0	2.0	15801.57	.638
36	9	3651.09	9334.59	2561.81	6953.50	1.0	2.0	15861.24	.641
54	3	7364.54	11619.13	7610.02	15721.11	.5	1.0	9475.67	.383

PROB. NO. M-8(01)

ATTACHMENT NO. 2

SHT. NO. 12 OF 14

Added by: Siva P. Dingam date: 1-30-85

ch'kd by: Randall's date: 1-31-85

EQUATION 9 - PRIMARY STRESSES

CONDITION . . . X-Y DBE, DEAD WT., PRESSURE  
EQ 9  
MOMENTS IN FT-LBS.

STRESSES IN PSI.

JOINT TYPE	HX	HY	HZ	M(X)	B1	B2	PRIMARY STRESS	RATIO TO ALLOWABLE	ALLOWABLE	
									2.25 Sm	
1 1	2318.32	12877.89	33757.88	33424.98	.5	1.0	12580.30	.297	< 37,575	KSI
2 9	2482.40	7482.58	23461.15	24750.29	1.0	2.0	22064.57	.521	< 37,575	KSI
3 9	2318.32	3587.92	41867.45	42024.81	1.0	2.0	28106.76	.664	"	"
5 9	2024.62	7120.80	11374.37	13571.34	1.0	2.0	18167.98	.429	"	"
6 1	2318.32	5246.59	13197.06	14389.71	.5	1.0	9242.19	.218	"	"
7 9	2024.62	7358.11	17759.43	19217.53	1.0	2.0	20136.04	.475	"	"
11 1	9803.24	8767.98	12269.16	17986.47	.5	1.0	9872.94	.233	"	"
12 1	9188.59	10695.21	12269.16	18690.91	.5	1.0	9996.47	.236	"	"
13 9	4692.66	4855.58	12269.16	14004.64	1.0	2.0	18319.02	.433	"	"
14 9	5805.91	4745.36	11995.40	14146.26	1.0	2.0	18368.38	.434	"	"
15 9	2176.89	5534.58	12269.16	13634.61	1.0	2.0	18190.04	.429	"	"
16 9	2717.08	4745.36	22687.09	23259.08	1.0	2.0	21544.76	.509	"	"
18 1	7090.56	4745.36	13224.78	15738.15	.5	1.0	9478.66	.224	"	"
23 9	2318.32	6629.38	11813.46	13743.40	1.0	2.0	18227.96	.430	"	"
6 9	2461.35	6278.92	14193.89	15714.63	1.0	2.0	18915.06	.447	"	"
3 3	10560.31	17027.65	23895.17	31183.98	.5	1.0	12187.31	.288	"	"

PROB. NO. M-8(01)

ATTACHMENT NO. 2

SHT. NO. 13 OF 14

Added by: Siva P. Lingam Date: 1-30-85

ch'ed by: Kanjana Date: 1-31-85

A-10



EQUATION 9 - PRIMARY STRESSES

CONDITION . . . 21Y DBE, DEAD WT., PRESSURE  
EQ 9

MOMENTS IN FT-LBS. STRESSES IN PSI.

JOINT TYPE	HX	HY	MZ	H(I)	B1	B2	PRIMARY STRESS	RATIO TO ALLOWABLE	ALLOWABLE
1 1	5835.29	18966.50	13066.77	22195.79	.5	1.0	10611.10	.251	2.25 Sm 37.575 KSI
2 9	6777.76	15954.67	25707.34	31531.99	1.0	2.0	24428.42	.577	" "
3 9	5835.29	5219.98	11649.67	14036.16	1.0	2.0	18330.00	.433	" "
5 9	6879.53	6516.00	3684.99	10166.87	1.0	2.0	16981.31	.401	" "
6 1	5835.29	8501.46	9627.05	14106.93	.5	1.0	9192.61	.217	" "
7 9	6879.53	12120.27	15735.53	21778.59	1.0	2.0	21028.74	.496	" "
11 1	17704.10	11409.37	3996.96	21437.92	.5	1.0	10478.20	.247	" "
12 1	20483.15	12007.76	3996.96	24077.41	.5	1.0	10941.07	.258	" "
13 9	13398.91	6361.97	3996.96	15360.90	1.0	2.0	18791.76	.444	" "
14 9	10423.35	8419.97	5193.87	14370.75	1.0	2.0	18446.63	.436	" "
15 9	4288.30	7278.01	3996.96	9345.30	1.0	2.0	16694.94	.394	" "
16 9	4423.30	8419.97	8052.70	12462.26	1.0	2.0	17781.40	.420	" "
18 1	9155.99	8419.97	11124.49	16585.99	.5	1.0	9645.58	.228	" "
23 9	5835.29	8567.60	4569.77	11326.60	1.0	2.0	17386.25	.410	" "
36 9	5999.27	9253.94	4190.83	11797.87	1.0	2.0	17549.82	.414	" "
54 3	11665.43	20521.87	11000.78	26042.64	.5	1.0	11285.70	.266	" "

PROB. NO. M-8(01)

ATTACHMENT # 2

SHT. NO. 14 OF 14

Added by: Siva P. Singam date: 1-30-85

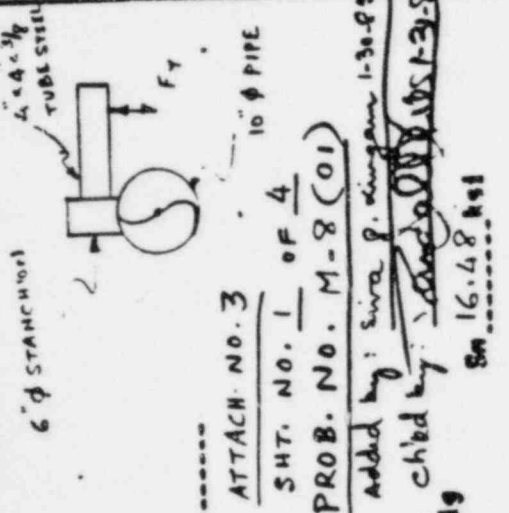
ch'ed by: Vandana date: 1-31-85

A-11



# CALCULATION SHEET

ORIGINATOR Siva P. Kingam DATE 11-30-84 CALC. NO. PS/PSU-1 REV. NO. 1  
 PROJECT DAVIS-BESSE UNIT 1 CHECKED [Signature] DATE 12-5-84  
 SUBJECT DATA SHEET FOR LOCAL STRESS CHECK SHEET NO. 11



FOR LONG TLF11

TO BE FILLED IN BY STRESS GROUP:

STRESS PROBLEM NO. M-8 ISSUE: C2 DATA POINT 40  
 PIPE SUPPORT NO. PSU-R1  
 TYPE OF SUPPORT: VERTICAL  (SNURBER)  Z  SHEU  ANCHOR

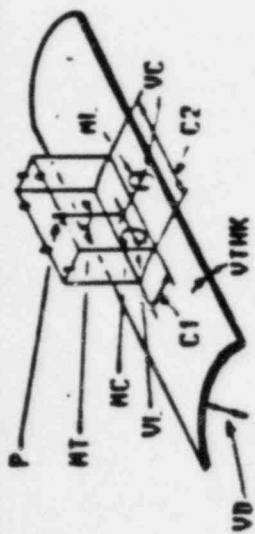
ATTACH. NO. 3 SHT. NO. 1 OF 4  
 PROB. NO. M-8 (01)  
 Added by: Siva P. Kingam 1-30-85  
 In ch'd by: [Signature] 1-24-85  
 PIPE WALL THK. 1.0 in  
 DESIGN PRESSURE 2500 psig  
 TEMPERATURE 670 °F  
 PIPE OUTSIDE DIAMETER 10.75 in  
 OPERATING PRESSURE 2750 psig  
 PIPING MATERIAL S.S. SA376-TP316  
 Sm 16.48 ksi

ME-101 LOAD CASE	SUPPORT LOADS						PIPE STRESS (psig)
	FORCES (lbs)			MOMENTS (in-lbs)			
	X	Y	Z	X	Y	Z	
MECHANICAL	-	-	-	-	-	-	825
WIND	2400	-	-	-	-	-	4415
SEISMIC	3900	-	-	-	-	-	8203
THERMAL	-	-	-	-	-	-	4546
RAIN	-	-	-	-	-	-	-

NOTE: 1) The stresses are at the center line of pipe.  
 2) The loads are as shown in figure.  
 • KSI FOR ME 210 IMPACT

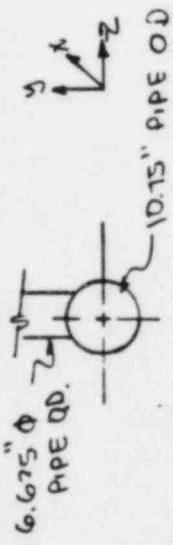


# CALCULATION SHEET

CALC. NO. PS/PSU-1 REV. NO. 1ORIGINATOR CH [signature] DATE 11/30/84 CHECKED T. Lin DATE 12/11/84PROJECT VA-19 PESSE UNIT 1 JOB NO. 12501SUBJECT NE 210 INPUT DATA SHEET NO. 121  
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50-> TYPE CALL 81A566+PROG.210IN1> TITLE, SUPPORT NUMBER PSU-01  
2> NUMBER OF LOADING CASES 3

### LOADING COMBINATIONS

CASE NUMBER	TYPE OF COMBINATION	P lbs	ML in-lbs	MC in-lbs	MT in-lbs	VL lbs	VC lbs	COMMENTS
1	WEIGHT+ DR	2400 ;	0 ;	20400 ;	0 ;	0 ;	0	
2	WEIGHT+ SSE	3900 ;	0 ;	33150 ;	0 ;	0 ;	0	
3	WEIGHT+THRM+ DR+SAH	2400 ;	0 ;	20400 ;	0 ;	0 ;	0	



### GEOMETRY

PIPE WALL THICKNESS EFFECTIVE IN	PIPE DIAMETER ACTUAL MEAN ADJUSTED IN	ATTACHMENT DIMENSIONS		SHAPE
		C1 in	C2 in	
A 1.0 ;	13.25 ;	6.625 ;	6.625 ;	1.0
B ;	;	;	;	;

GEOMETRIC PARAMETERS  $\beta$  AND  $\beta'$  CHECK

$\beta = C1/VD = .5$

$\beta = C2/VD = .5$

$\beta' = \frac{VD}{1.2 \times VTHK} = 6.625$

$0.01 < \beta < 0.5$  IF MT,  $5 < \beta' < 300$  ADJUST VB.

FOR POINT B:

$VT = \sqrt{VT^2 + VTHK^2}$

=

ATTACH. NO. 3SHT. NO. 2 OF 4PROB. NO. M-8(01)7> TYPE CALL 81A566+PROG.END  
-> TYPE PXOT AND 81A566+210RUM.-> TYPE CALL 81A566+PROG.210CUP

ADJUST PAPER, HIT RETURN

Added by: Siva P. Dingam 1-31-84  
checked by: [signature] 1-31-84

ORIGINATOR C.H. Ober...  
 PROJECT DAVIS RESE...  
 UNIT 1

DATE 11/30/84  
 JOB # 12501

CHECKER T. Lin DATE 12/11/82  
 STANDARD COMPUTER PROGRAM ME-210

CALC NUM. PS/PSU-1  
 REV NUM. 1

SHEET 13 OF

VERSION : 5  
 RELEASED : DEC 15, 1982  
 USER MANUAL VERSION : 1  
 THEORETICAL MANUAL VERSION : 1  
 VERIFICATION MANUAL VERSION : 5

B I J L A A R D S T R E S S A N A L Y S I S F O R C Y L I N D E R S

PSU-R1

ATTACH. NO. 3

SHT. NO. 3 OF 4

PROB. NO. M-8 (01)

I N P U T D A T A

	P (LBS)	ML (IN-LBS)	MC (IN-LBS)	MT (IN-LBS)	VL (LBS)	VC (LBS)	MA (FT-LBS)	MB (FT-LBS)
1	2400.	0.	20400.	0.	0.	0.	0.	0.
2	3900.	0.	33150.	0.	0.	0.	0.	0.
3	2400.	0.	20400.	0.	0.	0.	0.	0.

Added by: Siva P. Aringam 1-30-85

	VESTHK (IN)	VESDIA (IN)	C1 (IN)	C2 (IN)	SHAPE	PRESSURE (PSI)	SM (KSI)	BEND R (IN)
1	1.0000	13.2500	6.6250	6.6250	CIRCULAR	.0000	.0000	.0000
2	1.0000	13.2500	6.6250	6.6250	CIRCULAR	.0000	.0000	.0000
3	1.0000	13.2500	6.6250	6.6250	CIRCULAR	.0000	.0000	.0000

ch'ed by: T. Lin 1-31-85

MAXIMUM PRIMARY PLUS SECONDARY STRESS INTENSITY

1.0000      -1.18      .63      -1.18      .63      -5.06      3.87      2.71      -2.61

MAXIMUM PRIMARY PLUS SECONDARY STRESS INTENSITY

1.0000      -1.91      1.03      -1.91      1.03      -8.21      6.30      4.40      -4.25

MAXIMUM PRIMARY PLUS SECONDARY STRESS INTENSITY

1.0000      -1.18      .63      -1.18      .63      -5.06      3.87      2.71      -2.61



ORIGINATOR C.H. Ober  
 PROJECT DAVID BESE  
UNIT 1

DATE 11/30/84  
 JOB # 12501

CHECKER T. Liu DATE 12/1/84  
 STANDARD COMPUTER PROGRAM ME-210

CALC NUM. PS/PSU-1  
 REV NUM. J... SHEET - 14 OF

\*\*\*\*\* ME-210 POST-PROCESSOR (VERS. B) 12/8/81 \*\*\*\*\*

PROB. NO. M-8(01)  
ATTACH. NO. 3  
SHT. NO. 4 OF 4

DO YOU WANT TO ENTER PRESSURES AND OTHER STRESSES (YES OR NO)? >YES  
 ENTER ACTUAL PIPE O.D. AND WALL (INCHES) O.D.,W? >10.75,1.0  
 ENTER OPERATING AND DESIGN PRESSURE (PSIG) P-OP.,P-DES.? >2750,2500  
 IS ALL DATA O.K. SO FAR (YES OR NO)? >YES  
 ENTER FOLLOWING STRESSES WT, DBE, SSE, THRM, SAM (KSI)? >.825,4.415,8.203,4.546,0.0  
 ENTER SM STRESS ALLOWABLE (KSI)? >16.48  
 IS ALL DATA O.K. (YES OR NO)? >YES  
 PIPE O.D. = 10.75  
 PIPE WALL = 1.000  
 DESIGN PRESSURE = 2500  
 OPERATING PRESSURE = 2750  
 WEIGHT STRESS = .825  
 DBE STRESS = 4.415  
 SSE STRESS = 8.203  
 THERMAL STRESS = 4.546  
 SAM STRESS = 0.000  
 ALLOWABLE STRESS (SM) = 16.480

Added by: Siva P. Arigam 1-30-85  
 ch'kd by: Randall 1-30-85

CHECK VALUE ALLOWABLE

CHECK	VALUE	ALLOWABLE	
1A	14.03	24.720	upset primary in circumferential direction faulted primary " " upset primary in longitudinal direction faulted primary " " OK
1B	14.40	49.440	
2A	12.97	24.720	
2B	17.39	49.440	
3A	5.06 + 14.78 =	< 49.440	
3B	5.06 + 17.18 =	< 49.440	

FOR CHECK 3A AND 3B YOU MUST ADD THE MAX PRI + SEC STRESS INTENSITY FROM THE ME-210, RUN FOR LOAD CASE #3 AND MUST STILL BE BELOW THE SPECIFIED ALLOWABLE

Annex 54-371

Code Case N-411 Alternative Damping Values for Seismic Analysis of Piping  
Section III, Division 1 Class 1, 2, and 3.

Question:

What alternatives to the damping values given in Table N-1230-1, Appendix N,  
Section III Division 1 are acceptable for use in seismic analysis of Class 1, 2  
and 3 piping?

Reply:

It is the opinion of the Committee that for Section III, Division 1, Class 1, 2,  
and 3, construction, the damping value for seismic analysis of piping shown in  
Figure 1 may be used as an alternative to those given in Table N-1230-1,  
Appendix N.

The damping value in Figure 1 is applicable to both OBE and SSE, and is  
independent of pipe diameter.

This Code Case number shall be shown in the documentation for this analysis and  
on the Code Data Report.

PROB. NO. M-8(01)  
ATTACH. NO. 4  
SHT. NO. 1 OF 2

Added by: Siva P. Aringam date: 1-30-85

ch'kd by: Rangaswami date: 1-31-85

PROB. NO. M-8 (01)

ATTACH. NO. 4

SHT. NO. 2 OF 2

Added by: Uma P. Kingam 1-30-85

Checked by: [Signature] 1-31-85

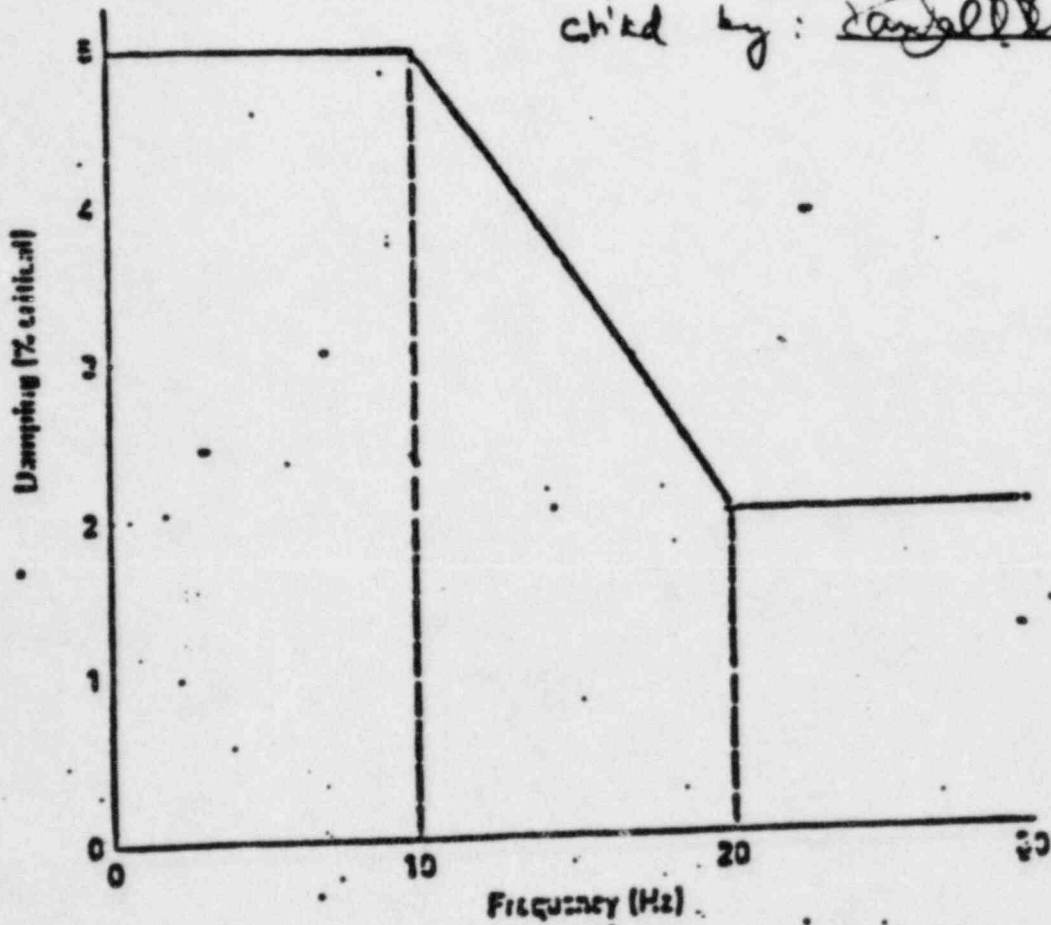


Figure 1

Damping Value for Seismic Analysis of Piping  
(Applicable to both OBE & SSE, Independent of Pipe Diameter)

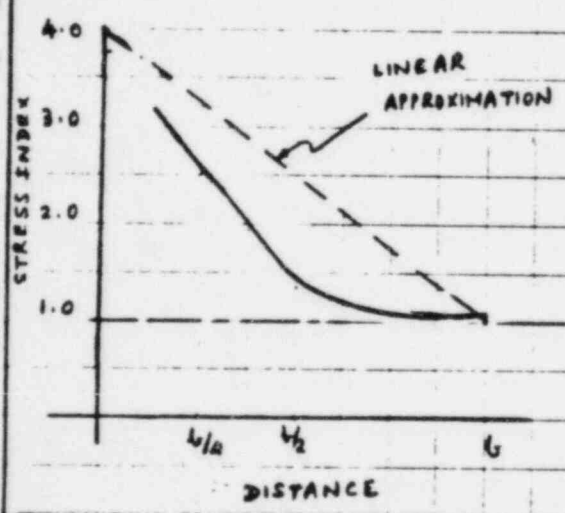


# CALCULATION SHEET

ATTACHMENT # 5

JOB NO. 12501 DAVIS-BESSE, UNIT-1		CALC. NO. M-8		REV. NO. 01	SHEET NO. 1 OF 1
PRESSURIZER SURGE LINE					
ORIGINATOR	DATE	CHECKED	DATE		
Siva P. Lingam	1-30-85	[Signature]	1-31-85		

1) ST. VENANT'S PRINCIPLE STATES THAT AT A DISTANCE EQUAL TO OR GREATER THAN THE MEMBER WIDTH (WALL THICKNESS IN THE CASE OF A PIPE) THE MEMBER STRESSES ARE UNAFFECTED BY LOAD APPLICATION OR GEOMETRIC DISCONTINUITY. REFERENCING WELDING RESEARCH COUNCIL (WRC) BULLETIN 19P, THE INTERFERENCE DISTANCE TO "ANY OTHER WELD OR DISCONTINUITY" IS DEFINED AS  $\sqrt{2r}$ . SINCE  $\sqrt{2r} = 2.208"$  IS GREATER THAN  $r = 1.0"$ , WE EVALUATED THE EFFECT OF ADJACENT DISCONTINUITY OUT TO A DISTANCE OF 2.208"



PLOT OF STRESS INDEX VS. DISTANCE FROM SOURCE  
(REF.: BEER & JOHNSON - "MECHANICS OF MATERIALS", PAGE 79)

AT  $l$ ,  $\frac{\sigma_{max}}{\sigma_{av.}} = 1.027$   
 AT  $\frac{l}{2}$ ,  $\frac{\sigma_{max}}{\sigma_{av.}} = 1.387$   
 AT  $\frac{l}{4}$ ,  $\frac{\sigma_{max}}{\sigma_{av.}} = 2.575$  WHERE  $l =$  MEMBER WIDTH

AS CAN BE SEEN, A LINEAR DISTRIBUTION OF STRESS INDEX AWAY FROM THE POINT OF STRESS CONCENTRATION IS VERY CONSERVATIVE.

2) USE OF CLASS-II FORMULA IN CLASS-I ANALYSIS CAN BE JUSTIFIED BY THE FOLLOWING FORMULA:

$$B, C, K \text{ or } i = \frac{\sigma}{s} = \frac{\text{elastic stress}}{\text{nominal stress}}$$

(Ref: D-101 of Appendix-D)