

ATTACHMENT NO. 2

TO

AEP:NRC:0514W

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**WHITING CORPORATION**  
 PRODUCTION ENGINEERING DEPT.  
 HARVEY, ILLINOIS 60426 U.S.A.  
 AREA CODE 312 331-4000

CUSTOMER AMERICAN ELECTRIC POWER  
 C.O. NO. C6884 REQN. 79604  
 DATE 8-21-87 BY MJM  
 PAGE i RI OF iii

Rev 1 MJM/ASZ 9-15-87 chkd RGG *MJM RGG*  
 (PP 3-7, 3-22, 4-111, A-1, C-3)

Attachment No. 2 to AEP:NRC:0514W

CRANE SEISMIC REPORT  
 CASK HANDLING CRANE  
 150 TON CAPACITY

EXISTING BRIDGE, S/N 10038  
 NEW TROLLEY S/N 12124

CUSTOMER: AMERICAN ELECTRIC POWER CORP.  
 COLUMBUS, OHIO  
 INDIANA & MICHIGAN ELECTRIC CO.  
 FOR: DONALD C COOK FACILITY

BRIDGMAN, MICHIGAN

P.O. NO.: C6884

SPECIFICATION: DCC-MB-105-QCN Rev. 0  
 DCCNE-101-QCN Rev. 0

**Charles R. Norman**  
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8-21-87ABSTRACT

The equipment reviewed in this report is an 'Electric Overhead Crane.' The crane is designed and rated for a capacity load of 150 tons on the main hook.

The crane was analyzed for the resistance to the specified Operational Base Earthquake (OBE) and the specified Safe Shutdown Earthquake (SSE). This was done with a load of 50 tons and no load on the main hook with the trolley at mid-span, quarter span, and end of span.

The crane was mathematically modeled as a multi-degree of freedom system of node points, interconnected by various finite elements. "ANSYS", a large scale general purpose computer program was used to perform a static and a reduced modal analysis. It was found that excitations parallel to the runway (Y direction) would produce slip. This excitation was then proportioned to produce a maximum Y reaction that would not produce slip. Those components not directly analyzed by the computer program were manually analyzed with loadings from the computer program.

It was found that the stresses in the principal structural components did not exceed the allowable stresses with a 50 ton load on the main hook.

WHITING REQN. 79604 DATE 8-21-87  
BY MJM PAGE iii OF iii  
*CWJ 8-27-87*

TABLE OF CONTENTS

SECTION

- 1 ANALYSIS DESCRIPTION
- 2 SUMMARY OF RESULTS
- 3 GEOMETRY SECTION
- 4 SUPPLEMENTAL CALCULATIONS

APPENDICIES

- A EXCITATION AND RESPONSE SPECTRA,  
NATURAL FREQUENCIES AND MODE COEFFICIENTS
- B SUMMARY OF COMPUTER RESULTS  
CRANE STRESSES, REACTIONS & COMPONENT LOADINGS
- C NOMENCLATURE, REFERENCES
- D EVALUATION OF 55 TON LOAD

WHITING REQN. 79604 DATE 8-21-87  
BY MJM PAGE 1-1 OF 4  
*8/29/87*

### ANALYSIS DESCRIPTION

The crane was analyzed to determine the effect of seismic excitations. For this analysis, the matrix displacement method was used based upon finite element techniques. The crane was mathematically modeled as a system of node points interconnected by various finite elements representing straight beams. All masses and inertias were distributed among the nodes whose degrees of freedom characterize the response of the structure. The interconnecting finite elements were assigned stiffnesses equivalent to that of the actual structure.

The mathematical model represents as accurately as possible the flexibility of the bridge girders, hoist rope, and girder end connection. The trolley, the drive units and the bridge trucks were represented as rigid bodies.

The crane was analyzed with the trolley positioned at mid-span, quarter span and end of span. This was done with a load of 50 tons in the high and low position. The crane was also analyzed with an unloaded trolley at mid-span, quarter span and end of span.

The dynamic analysis was of the mode frequency (MODAL) type, solving for the resonant frequencies and the mode shapes that characterize the crane. The modes with meaningful participation in a given direction are directly expanded by the computer program to yield the expanded mode shapes, the element stresses and the reaction values. This type of analysis is linear and plastic deformation, sliding, friction, and slack rope are not taken into account.

The amplified response spectra used in the analysis are shown in Appendix 'A'. These include the three orthogonal excitations for the specified earthquakes. Also included in this Appendix are the mode coefficients and natural frequencies for mode shapes considered.

WHITING REQN. 79604 DATE 8-21-87  
BY MJM PAGE 1-3 OF 4  
*copy 8-27-87*

Impact factors for wheel flange to rail contact, etc., have been considered negligible. The state of the art today is such that these impacts cannot rigorously be studied; however, independent time history analyses have been run in many cases, all indicating slow relative motion between the rail and the wheel. This is because of the time dependency of the forcing function coming from the building into the crane. Note that the only coupling through which these forces can be transmitted is dynamic friction. Upon reaching the rail the wheel will first rise through the corner radius and then contact the rail. During this period, the structure is starting to deflect as the end of the crane in this direction is flexible.

The computer analysis was performed using ANSYS, a large scale finite element program.

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WHITING REQN. 79604 DATE 6-11-87  
 BY MJM PAGE 1-4 OF 4  
*6-11-87*

TABLE 1-1

## MODE COEFFICIENT SIGNIFICANCE

CASES	TROLLEY	EXCITATION DIRECTION		
		X	Y	Z
LOADED	MID	.005	.0005	.0005
	1/4	.005	.001	.001
	END	.005	.002	.002
NO LOAD	MID	.005	.0005	.001
	1/4	.005	.001	.002
	END	.005	.002	.004

WHITING REQ. 79604 DATE 8-26-87  
BY MJM PAGE 2-1 OF 8  
*WJW 8-27-87*

### SUMMARY OF RESULTS

The crane was mathematically modeled using finite elements. On the basis of preliminary runs, the number of degrees of freedom and the significance criteria for modal expansion were adjusted. Static and three load step reduced modal runs were made and the results summed. Because slip occurs, the y excitation was proportioned and these results summed again.

The crane was analyzed with the new trolley at mid-span, quarter span and end of span. For these positions the analysis was done with a 50 ton load on the main hook and with no load.

Tables 2-1 and 2-2 summarize the maximum stresses in the members from the finite element model. Tables 2-3 and 2-4 summarize the maximum stresses from the manual calculations using the loadings from the finite element model. All stresses are within the allowables required by the job specification with a 50 ton load. Table 2-5 summarizes the buckling stability of the girder web which is also within the allowables required by the job specification.

Table 2-6 summarizes the rope load from the finite element model. Because of the seismic acceleration a slack rope condition was found to exist under certain conditions. This cannot be truly simulated with a linear modal analysis. However our experience with time history analyses shows that a modal analysis tends to produce conservative results. The rope load predicted by the modal analysis is well below the allowable rope load.

Table 2-7 summarizes the maximum crane bridge wheel loads. When the excess dynamic rope load (that which produces a slack rope) is deducted, a small upkick is produced by the loading conditions examined. When the wheel loads parallel to the runway are compared with the vertical wheel load times the coefficient of friction, it is found that the crane bridge will tend to slide under certain loading conditions examined. This sliding is oscillatory in nature and the loadings predicted by a modal analysis are conservative. The reported wheel loads have been adjusted to account for frictional effects.

Although some non-linearities are produced by the specified excitations the specified linear analysis will conservatively predict the behavior of the crane during a seismic excitation.

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WHITING REQN. 79604 DATE 8-26-87  
BY MJM PAGE 2-2 OF 8  
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Additional information on the response of the crane may be found in Appendix 'A'.

The crane was found to meet the job specification requirements for a seismic excitation with a 50 ton load on the main hook.

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WHITING REQN. 79604 DATE 6-19-87  
 BY MJM PAGE 2-3 OF 8  
02/18-27-87

TABLE 2-1 OBE  
 SUMMARY OF MAXIMUM STRESS  
 FROM COMPUTER OUTPUT

COMPONENT	TROLLEY	LOAD	ELEMENT	NODE	STRESS KSI	ALLOW KSI	<u>STRESS</u> <u>ALLOW</u>
GIRDER A	MID	50 DN	28	312	21.8	24.0	.91
GIRDER B	MID	50 DN	52	361	21.5	24.0	.90
END TIE RHE	MID	50 UP	17	154	13.4	24.0	.56
END TIE LHE	END	NO	75	254	13.3	24.0	.55

FOR ADDITIONAL DETAILS SEE TABLES B1 TO B18.

TABLE 2-2 SSE  
 SUMMARY OF MAXIMUM STRESS  
 FROM COMPUTER OUTPUT

COMPONENT	TROLLEY	LOAD	ELEMENT	NOTE	STRESS KSI	ALLOW KSI	<u>STRESS</u> <u>ALLOW</u>
GIRDER A	MID	50 DN	28	312	31.5	32.7	.96
GIRDER B	MID	50 DN	52	361	30.9	32.7	.94
END TIE RHE	MID	50 UP	17	154	20.3	32.7	.62
END TIE LHE	END	NO	75	254	23.6	32.7	.72

FOR ADDITIONAL DETAILS SEE TABLES B1 TO B18.

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WHITING REQN. 79604 DATE 8-26-87  
 BY ASZ PAGE 2-4 OF 8  
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TABLE 2-3 OBE

SUMMARY OF MAXIMUM STRESS FROM SUPPLEMENTARY

CALCULATION. (FOR ADDITIONAL DETAILS SEE REFERENCED PAGES)

COMPONENT	DETAIL	PAGE	STRESS (KSI)	ALLOW. (KSI)	STRESS ALLOW.
BRIDGE WHEEL	FLANGE SHEAR	4-14	1.3	21.2	0.06
BRIDGE AXLE	SHEAR	4-15	6.8	24.0	0.28
BRIDGE TRUCK SEISMIC LUGS:	LUG PLATE				
	SHEAR	4-18	1.1	14.4	0.08
	TENSION	4-19	5.5	24.0	0.23
	LUG PIN				
	SHEAR	4-22	1.5	12.0	0.13
	WELDS	4-24	3.8	20.4	0.19
	BOLTS TENSION	4-26	10.8	61.3	0.18
	BOLTS SHEAR	4-26	6.9	36.8	0.19
BRIDGE TRUCK	TENSION	4-29	11.5	24.0	0.48
	SHEAR	4-29	8.1	14.4	0.56
TRUCK TO GIRDER	BOLTS	4-46	28.1	36.8	0.76
	WELDS	4-61	7.9	20.0	0.40
GIRDER TO END TIE	BOLTS	4-79	28.7	36.8	0.78
	WELDS	4-88	14.0	20.0	0.70
GIRDER	WELDS	4-104	10.5	20.0	0.53
GIRDER END	SHEAR	4-105	10.2	14.4	0.71
	WELDS	4-105	12.0	20.0	0.60



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WHITING REQ. 79604 DATE 8-26-87  
 BY ASZ PAGE 2-5 OF 2  
 8.27.87

TABLE 2-4

SSE

SUMMARY OF MAXIMUM STRESS FROM SUPPLEMENTARY

CALCULATION (FOR ADDITIONAL DETAILS SEE REFERENCED PAGES)

COMPONENT	DETAIL	PAGE	STRESS (KSI)	ALLOW. (KSI)	STRESS ALLOW.	
BRIDGE WHEEL	FLANGE SHEAR	4-14	2.4	28.9	0.08	
BRIDGE AXLE	SHEAR	4-15	9.9	32.7	0.30	
BRIDGE TRUCK SEISMIC LUGS:	LUG PLATE	SHEAR	4-18	2.9	19.6	0.15
	LUG PLATE	TENSION	4-19	15.1	32.7	0.46
	LUG PIN	SHEAR	4-22	4.0	16.4	0.24
	WELDS	4-24	10.5	27.8	0.38	
	BOLTS TENSION	4-26	29.7	83.6	0.36	
	BOLTS SHEAR	4-26	19.0	50.2	0.38	
BRIDGE TRUCK	TENSION	4-33	16.8	32.7	0.51	
	SHEAR	4-33	11.8	19.6	0.60	
TRUCK TO GIRDER	BOLTS	4-52	46.2	50.2	0.92	
	WELDS	4-70	13.2	27.3	0.48	
GIRDER TO END TIE	BOLTS	4-84	47.7	50.2	0.95	
	WELDS	4-92	23.2	27.3	0.85	
GIRDER	WELDS	4-104	15.1	27.3	0.55	
GIRDER END	SHEAR	4-105	13.8	19.6	0.70	
	WELDS	4-105	16.3	27.3	0.60	

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WHITING REQN. 79604 DATE 8-26-87  
 BY ASZ PAGE 2-6 OF 8  
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## TABLE 2-5

SUMMARY OF BUCKLING STABILITY  
 FROM SUPPLEMENTARY CALCULATIONS.

COMPONENT		PAGE	STABILITY FACTOR	ALLOW. FACTOR	STABILITY ALLOW.
GIRDER WEB	OBE	4-103	0.597	0.667	0.90
	SSE	4-103	0.876	0.909	0.96

FOR ADDITIONAL DETAILS SEE REFERENCED PAGES.

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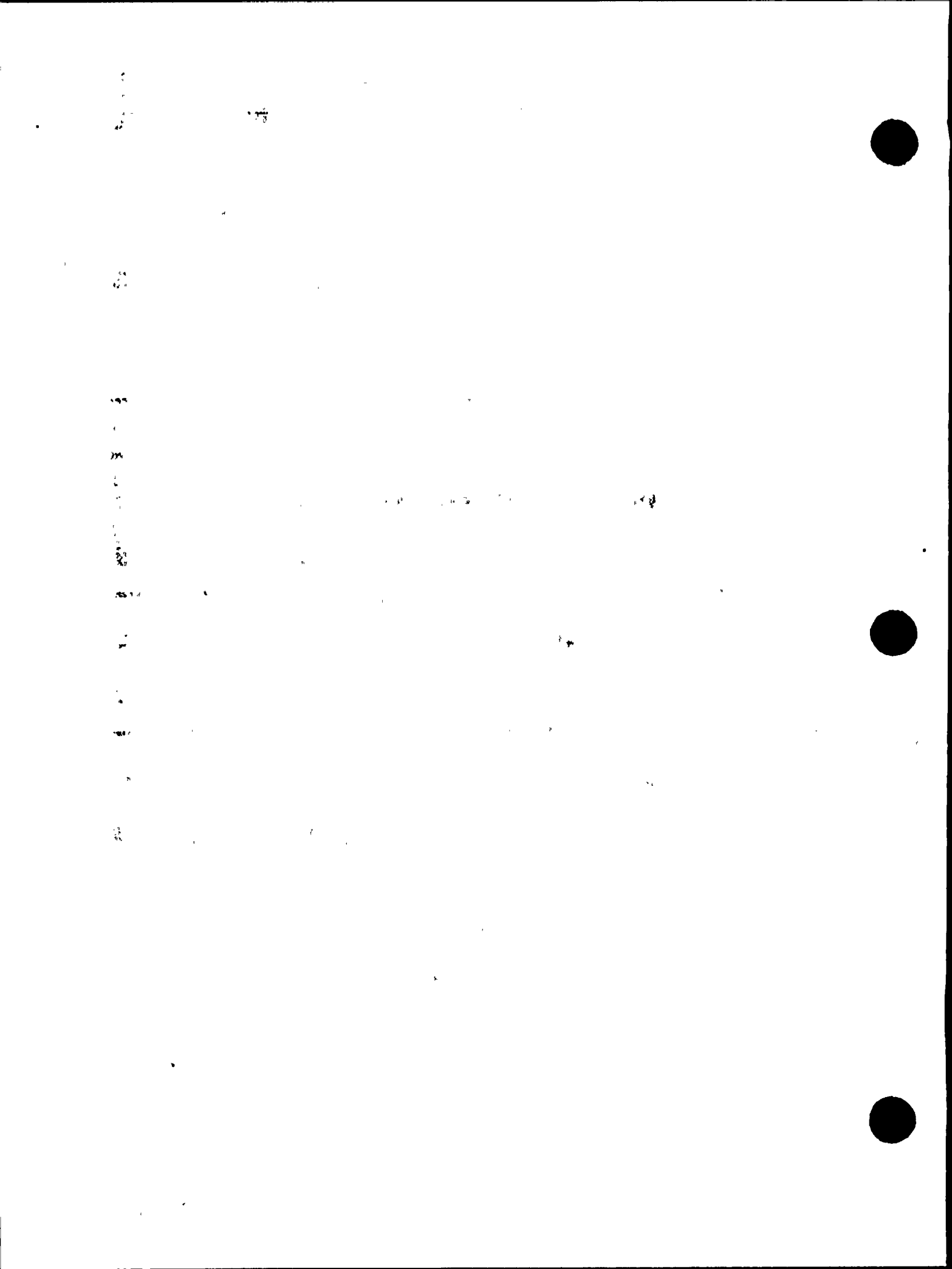
WHITING REQ. 79604 DATE 8-27-87  
 BY ASZ PAGE 2-7 OF 8  
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TABLE 2-6

SUMMARY OF MAXIMUM ROPE LOAD  
 (KIPS)

	TROLLEY	LOAD	STATIC	SUM	DIFFER.
OBE	MID.	DN	120.0	367.3	-127.3
SSE	MID	DN	120.0	583.7	-343.6

FOR ADDITIONAL DETAILS SEE TABLES B61 TO B64



WHITING REQ. 79604 DATE 2-27-87  
 BY ASZ PAGE 2-8 OF 8  
*copy 8.27.87*

TABLE 2-7

SUMMARY OF MAXIMUM WHEEL LOADS  
 (KIPS)

		$W_x$ MAX AT 1/4-DN	$W_y$ MAX AT END-DN	MAX. $W_z$		SEISMIC LUG LOAD
				$W_A$ MAX AT END-DN	$W_B^*$ AT 1/4-DN	
OBE	DRIVER 101,201	5.88	44.9	179.6	71.2	13.3 AT MID-DN
	IDLER 102,202	5.88	—	138.1	106.0	2.7 AT END-NO
OSSE	DRIVER 101,201	11.1	64.2	256.8	97.1	36.6 AT END-DN
	IDLER 102,202	11.1	—	195.2	147.1	15.3 AT END-NO

\*  $W_B$  IS LOAD ON OTHER WHEEL WHEN  $W_A$  IS MAX.

FOR ADDITIONAL DETAILS SEE TABLES 4-4 AND 4-5.



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WHITING REQN. 79604 DATE 8-21-87  
BY MJM PAGE 3-1 OF 24  
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### GEOMETRY SECTION

The equipment analyzed in this report is an 'Electric Overhead Crane' which is designed and rated for a capacity load of 150 tons on the main hook. This is based on using the new SFP design trolley (S/N 12124) on the existing bridge (S/N 10038) after recommended field modifications. For this analysis the lifted load is limited to 50 tons and a hook approach of 10'-3-1/4" when loaded.

The mathematical model of the crane with node numbering and global coordinates is illustrated on pages 3-7 through 3-10.

The boundary conditions tabulated in table 3-1 are selected to provide the most realistic linear approximation to actual conditions in a seismic event as follows:

NODES - 101, 102, 201, 202

UZ: Simulates wheel to rail contact in the vertical direction.

NODES - 101, 201

UY: Simulates the drive brake which is automatically set and which provides stability parallel the runway.

NODES - 101, 102, 201, 202

ROTX: Simulates the differential wheel loads of a fixed bogie truck subject to overturning.

NODE - 124

UX: Simulates wheel to rail contact perpendicular to the runway.

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WHITING REQ. 79604 DATE 8-21-87  
BY MJM PAGE 3-2 OF 24  
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The other restraints of nodes 123 and 124 were selected to simplify the analysis.

Those nodes which are coupled have the same displacement in the indicated directions only. Their displacements in all other directions are independent (released). This coupling is used to simulate load transfer between various components and is tabulated in table 3-2.

#### BRIDGE TRUCK

NODES - 101-121, 102-122

UX: Simulates the load transfer from the bridge wheels to the runway rail perpendicular the runway.

#### TROLLEY

NODES - 391-401, 392-402, 393-403, 394-404

UZ: Simulates wheel to rail contact in the vertical direction.

NODES - 393-403, 394-404

UX: Simulates the drive brake which is automatically set and which provides stability parallel the girders.

NODES - 392-402, 393-403

UY: Simulates wheel to rail contact perpendicular to the girders.

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WHITING REQ. 79604 DATE 8-21-87  
BY MJM PAGE 3-3 OF 24  
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The master dynamic degrees of freedom for a reduced modal analysis tabulated in table 3-3 are selected to obtain those modal shapes which characterize the principal vibrations of the structure. Placement is such as to include coupled modal shapes due to eccentricities. Higher degrees of freedom were not included because they will not contribute significantly to the system response. This can be justified by the responses obtained.

The girders, and the girder end connections are modeled as uniform beams. The rope is modeled as a spar element which is capable of supporting axial loads only. These elements have the properties of the corresponding parts of the actual crane. The trolley, the drive, and certain short connections are modeled as rigid members capable of transmitting loads only. The bridge trucks are simulated with a beam to simplify the wheel load analysis. Lumped masses were assigned to represent the masses of the trolley, the bridge trucks, the drive and the wheels. Additionally the beam members were assigned distributed masses.

The trolley and drives were modeled as rigid members because past experience shows that components of this type are very stiff structures with high natural frequencies in excess of 40 Hz.

The simulation of the restraint of the crane perpendicular to the runway is modeled on only one side consisting of a linear spring and two rigid beams capable of transmitting the load to the bridge wheels. The spring stiffness is selected so that the resulting frequency of the x mode yields an acceleration value from the high frequency region of the response spectrum curve. The resulting loads are distributed to the two runway rails by the 2/3, 1/3 method. The reason for the 2/3, 1/3 distribution is to account for manufacturing tolerances in which case one end of the crane would tend to contact the runway rail before the other end. The other end would however carry a portion of the reaction due to frictional resistance to sliding before flanging of the wheels.

Although certain simplifications are employed in making the linear mathematical model, these simplifications are in accordance with accepted practice. Such simplifications are employed to provide a model solveable with available resources while predicting the seismic response with reasonable accuracy.

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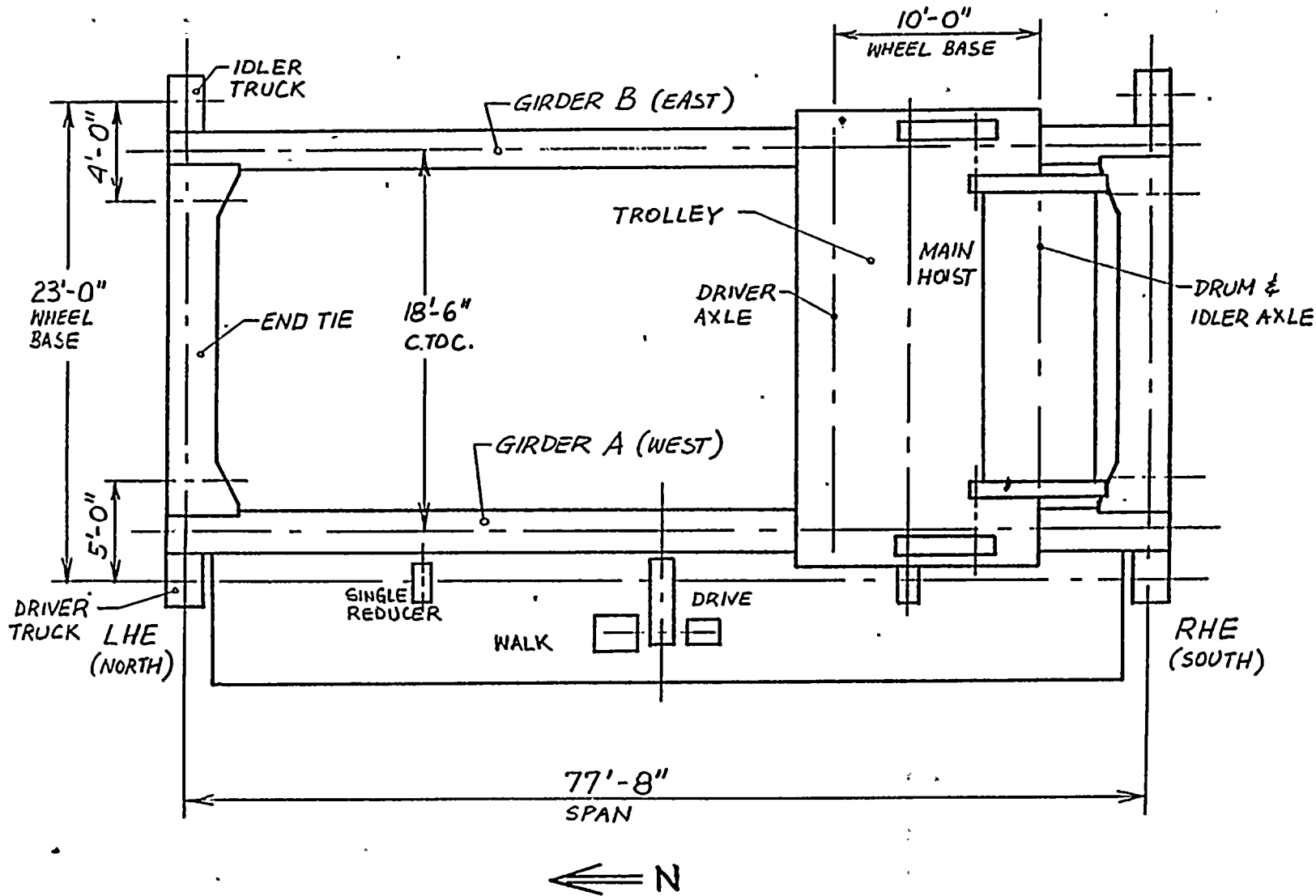
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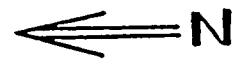
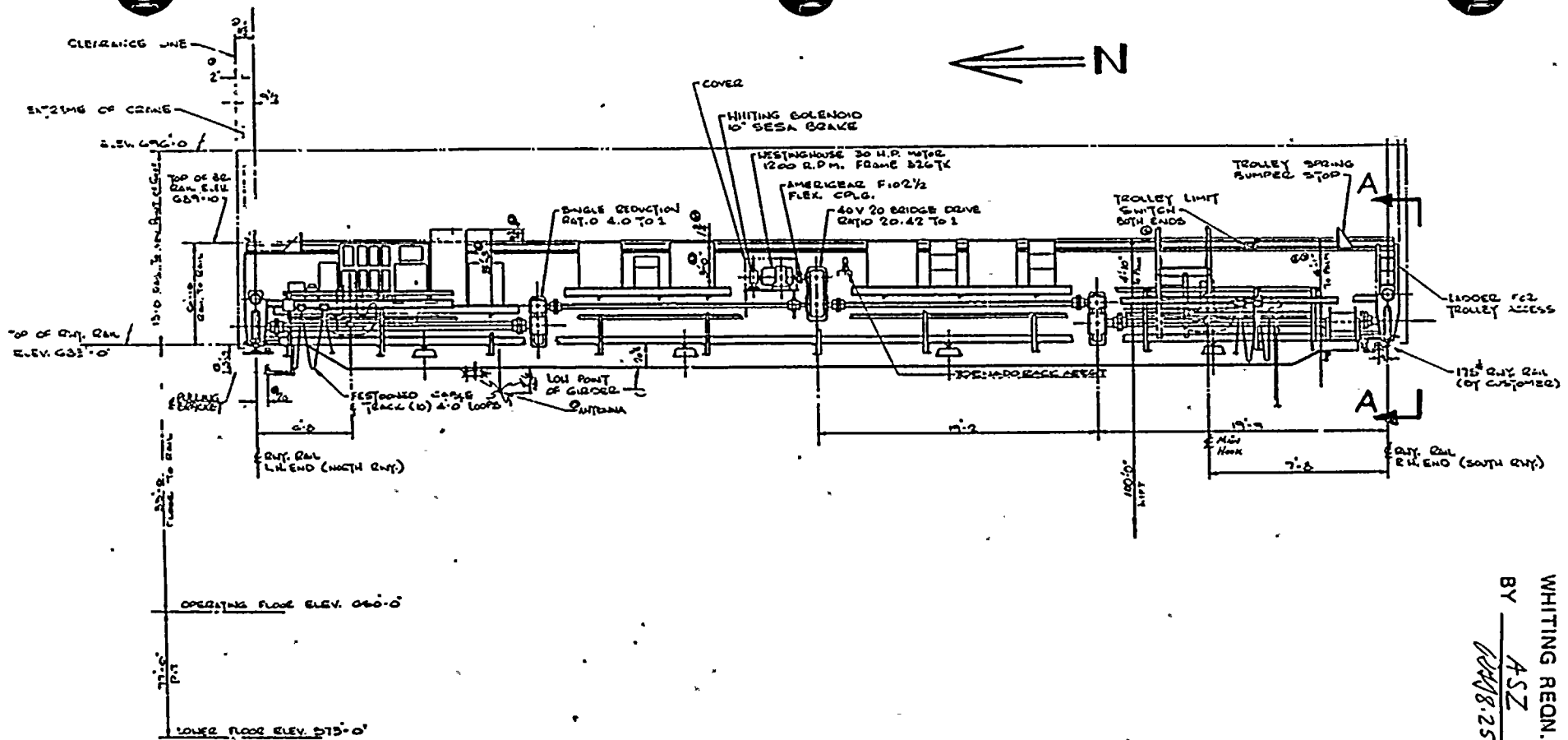


# CRANE PLAN VIEW



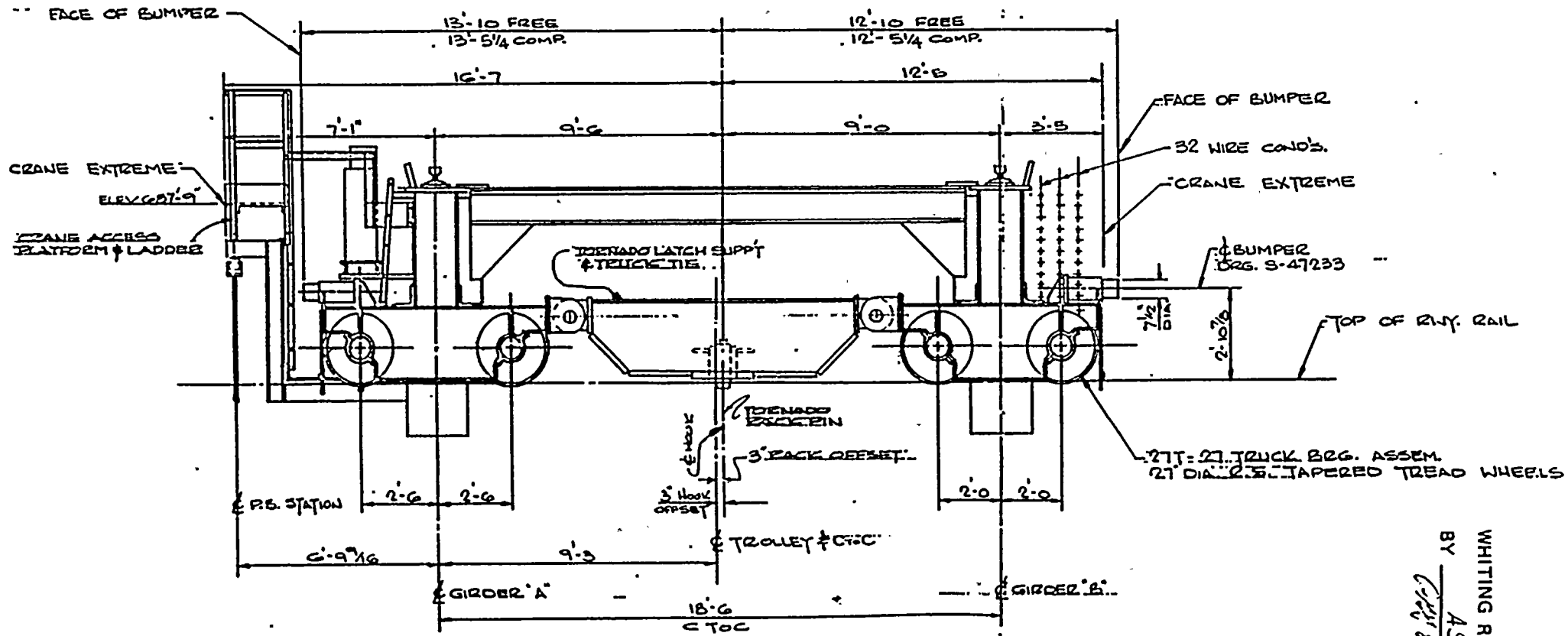
WHITING REQN. 79604 DATE 6-15-87  
 BY ASZ 8-25-87 PAGE 3-4 OF 24





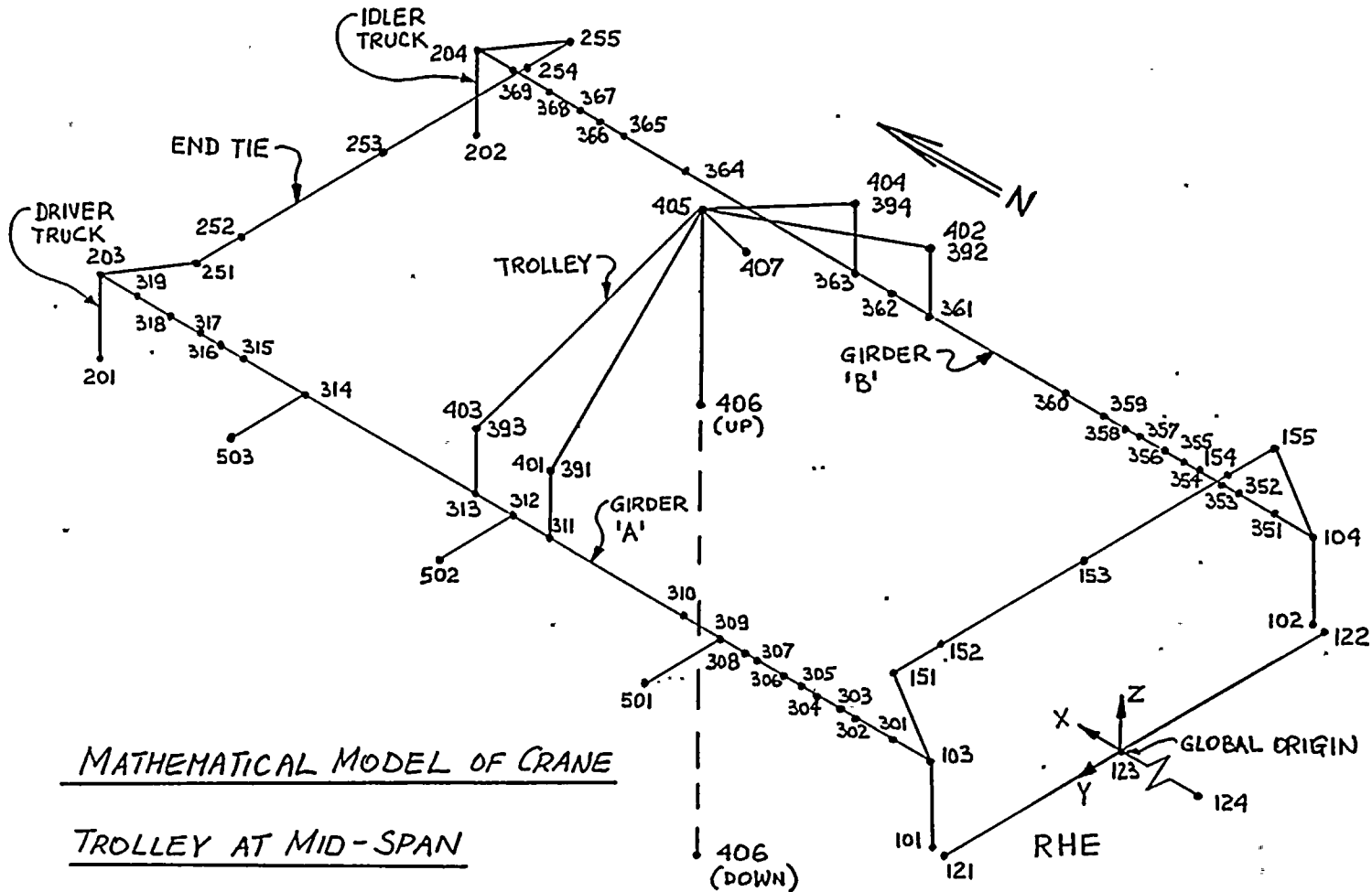
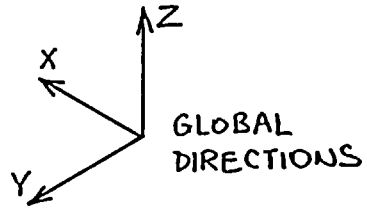
GENERAL ARRANGEMENT OF ONE MOTOR BRIDGE  
FRONT ELEVATION.

WHITING REQN. 79604 DATE 6-16-87  
 BY ASZ PAGE 3-5 OF 24  
08/8-25-87



--- VIEW 'A-A'  
 BRIDGE END ELEVATION

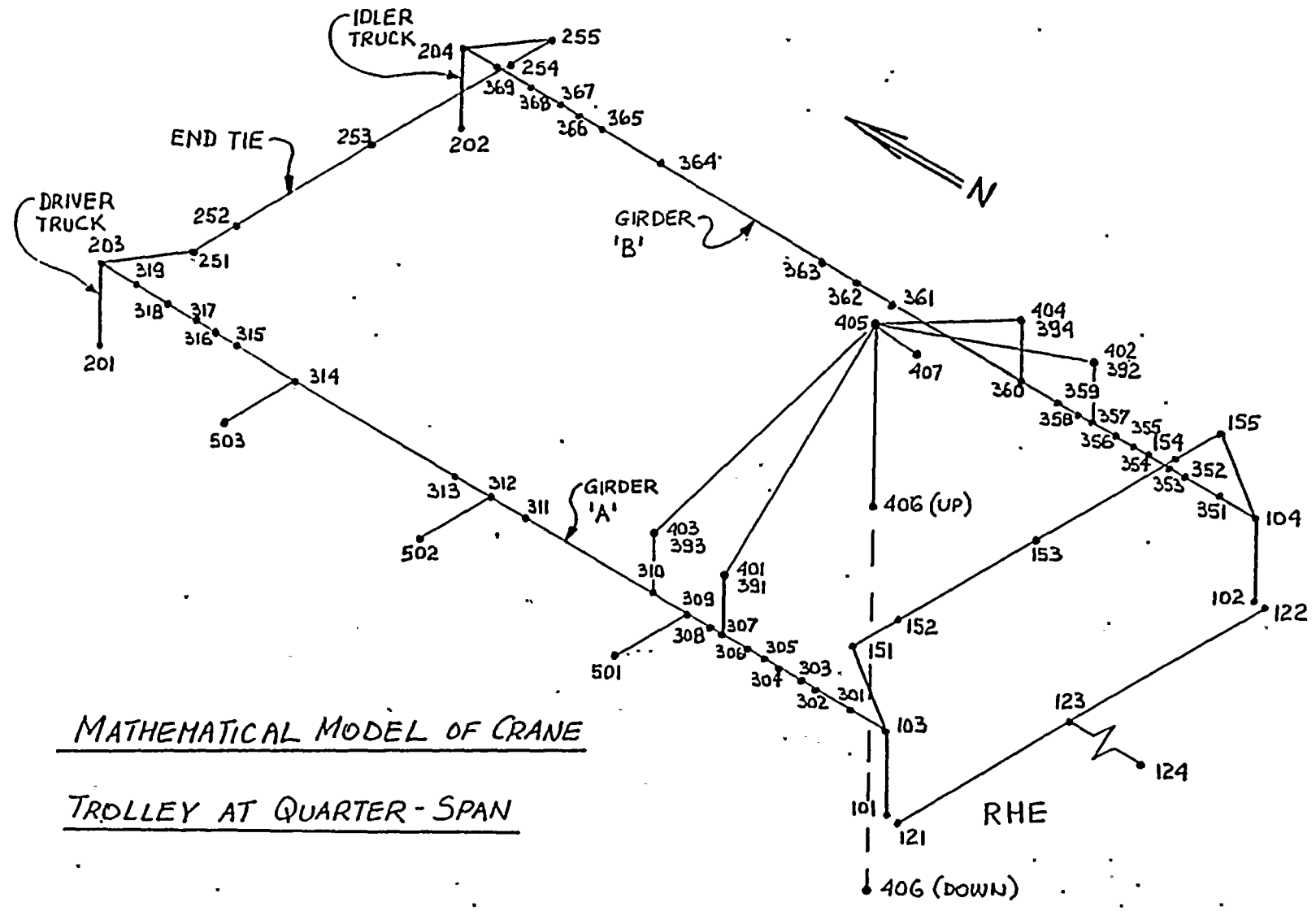
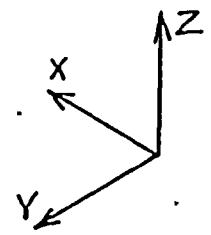
WHITING REQN. 79604 DATE 6-16-87  
 BY ASZ PAGE 3-6 OF 27  
 8/25/87



MATHEMATICAL MODEL OF CRANE

TROLLEY AT MID-SPAN

WHITTING REQN. 79 604 DATE 6-16-87  
 BY ASZ PAGE 3-7R1 OF 24  
 REV. ASZ 9-14-87 RWD 9/14/87



MATHEMATICAL MODEL OF CRANE

TROLLEY AT QUARTER-SPAN

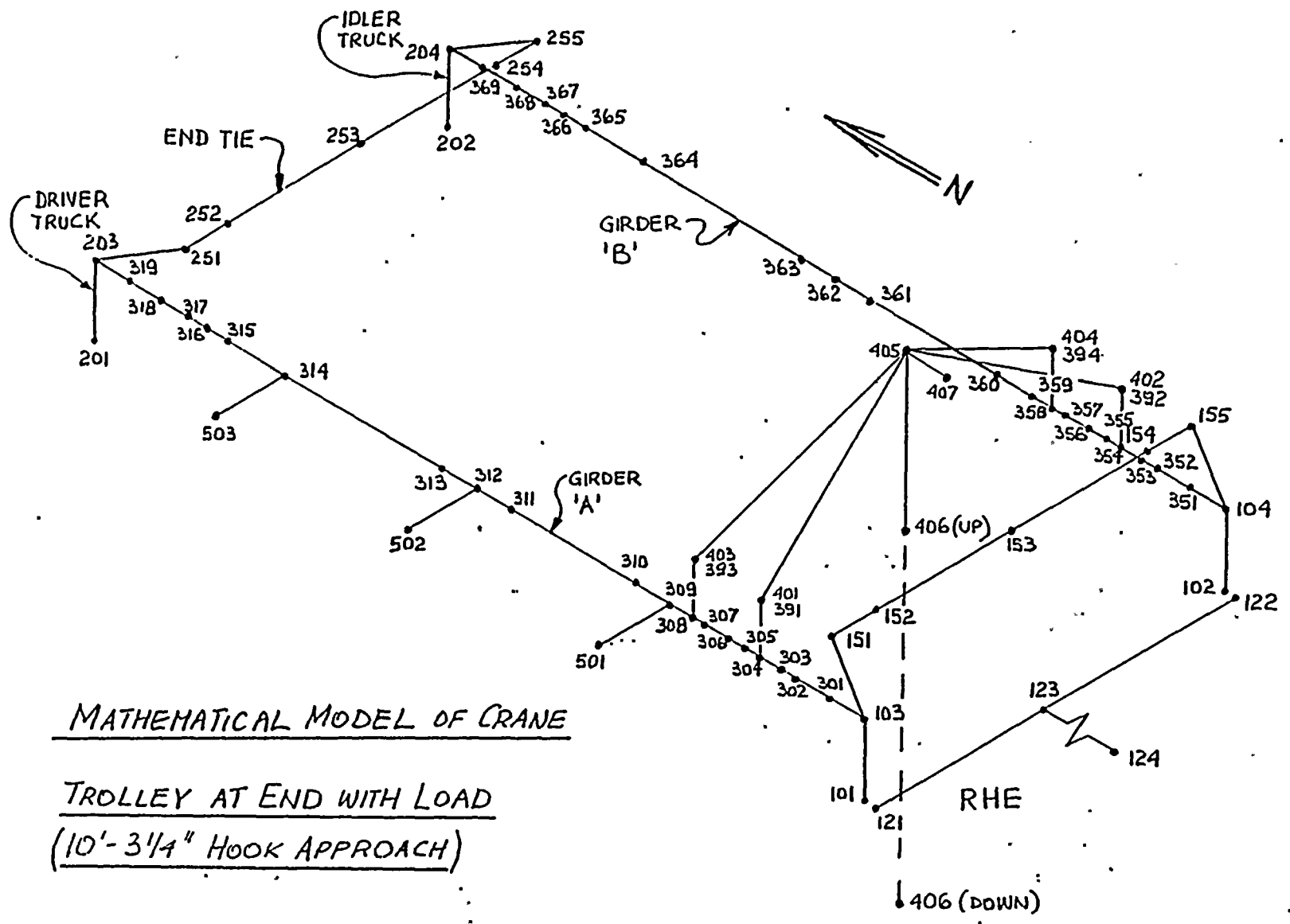
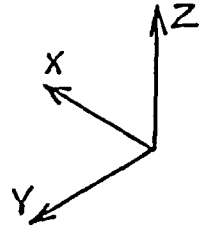
WHITING REQN. 79604 DATE 6-16-87  
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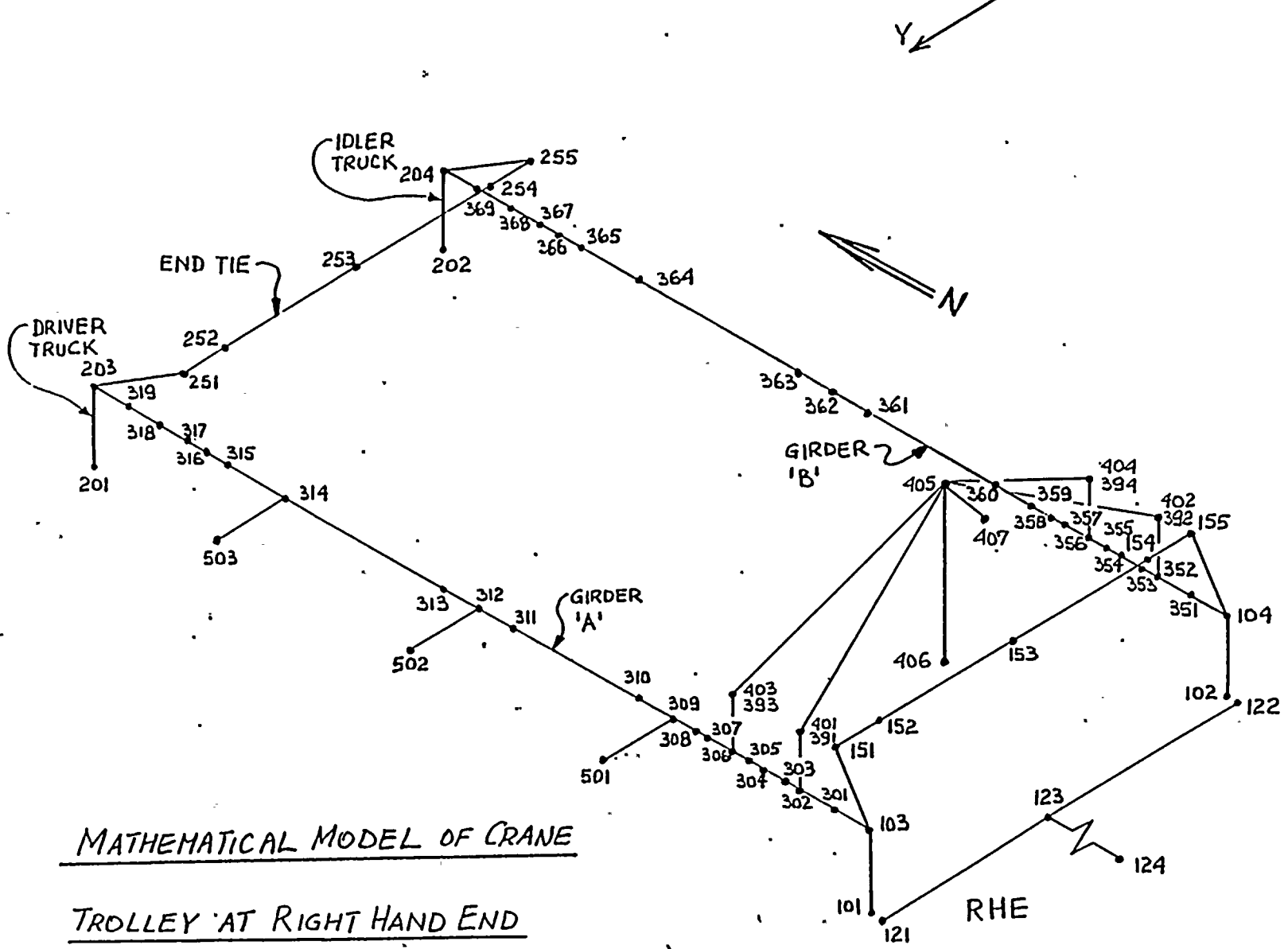
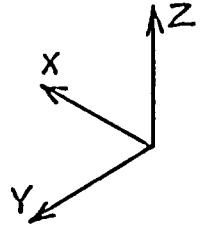
102





MATHEMATICAL MODEL OF CRANE  
TROLLEY AT END WITH LOAD  
(10'-3 1/4" HOOK APPROACH)

WHITING REQN. 79604 DATE 6-16-87  
 BY ASZ PAGE 3-9 OF 24  
*08/25/87*



MATHEMATICAL MODEL OF CRANE  
TROLLEY AT RIGHT HAND END  
WITHOUT LOAD (7'-8" HOOK APPROACH)

WHITING RECN. 79 604 DATE 6-16-87  
 BY ASZ PAGE 3-10 OF 27  
*Chap 82587*

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WHITING REQ. 79604 DATE 6-9-87  
 BY MJM PAGE 3-11 OF 24  
*6/25/87*

TABLE 3-1

BOUNDARY CONDITIONS  
 NODES WITH ZERO DISPLACEMENT IN THE INDICATED DIRECTIONS

LOCATION	NODE	TRANSLATION			ROTATION		
		UX	UY	UZ	ROTX	ROTY	ROTZ
BR SPRING	124	X	X	X	X	X	X
BR SPRING	123		X	X	X	X	
TRUCK, D, R	101		X	X	X		
, I, R	102			X	X		
TRUCK, O, L	201		X	X	X		
, I, L	202			X	X		

TABLE 3-2

COUPLED NODES  
 NODES WITH EQUAL DISPLACEMENT IN THE INDICATED DIRECTIONS

LOCATION	NODES		TRANSLATION			ROTATION		
			UX	UY	UZ	ROTX	ROTY	ROTZ
BR SPRING	101	121	X					
	102	122	X					
TROLLEY	401	391			X			
	402	392		X	X			
	403	393	X	X	X			
	404	394	X		X			

WHITING REQ. 79604 DATE 6-9-87  
 BY MJM PAGE 3-12 OF 24  
*0228-2587*

TABLE 3-3

## DYNAMIC DEGREES OF FREEDOM

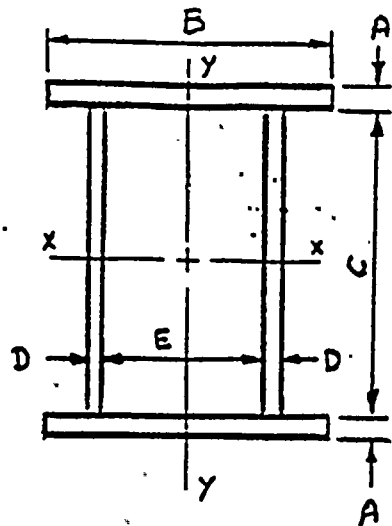
MASTER NODES FOR REDUCED MODAL ANALYSIS  
 FREE IN INDICATED DIRECTION

LOCATION	NODE	TRANSLATION		
		UX	UY	UZ
GIRDER A	309		X	X
	312	X	X	X
	314		X	X
GIRDER B	359		X	X
	362	X	X	X
	364		X	X
END CONN R	153	X		X
	L 253	X		X
TROLLEY	407	X	X	X
LOAD	406			X
DRIVE MACH	501	X		X
	502	X		X
	503	X		X

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SYMMETRICAL BOX GIRDER PROPERTIES



PROGRAM 107 PROGRAM ID 1-A-1-09(019)

WHITING REQ# 79604 DATE 6-8-87

BY MJM PAGE 3-13 OF 24

UWA 8-25-87

ORIGINAL GIRDER

.....  
107

	READ		ENTER
1.2500	-----	1	A <u>1.25</u>
21.0000	-----	2	B <u>21.</u>
93.0000	-----	3	C <u>93.</u>
0.3125	-----	4	D <u>.3125</u>
17.0000	-----	5	E <u>17.</u>

158490.6250	-----		I <sub>xx</sub>
3319.1753	-----		S <sub>xx</sub>
6285.1928	-----		I <sub>yy</sub>
598.5897	-----		S <sub>yy</sub>
110.6250	-----		AREA
16480.3619	-----		TORSIONAL CONSTANT (K)

.....

# GIRDER END

## SECTION PROPERTIES

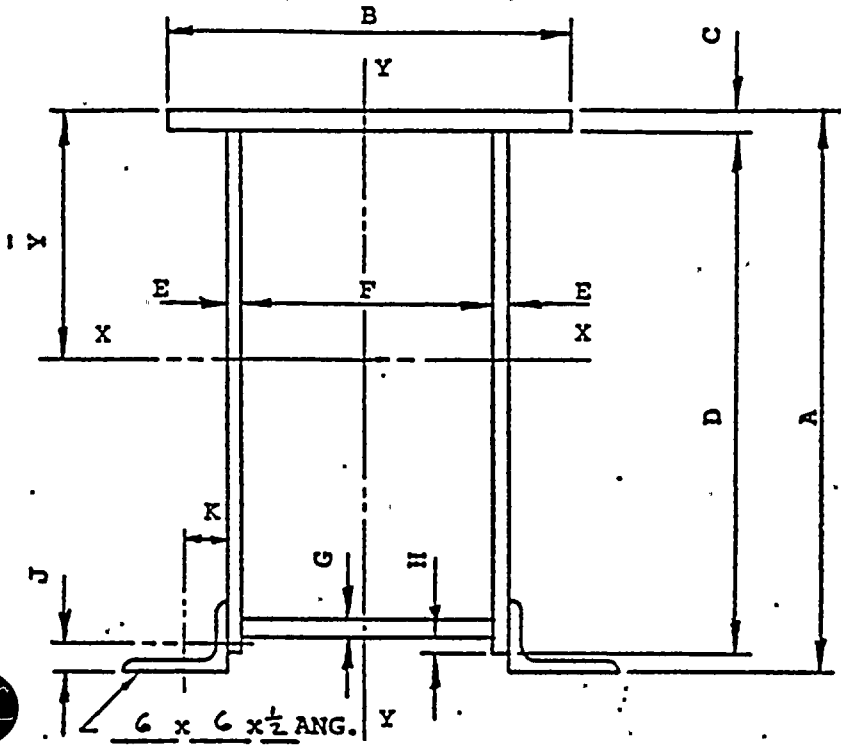
### SECTION "CC"

PROGRAM 117 PROGRAM ID 1-A-2-06 (046)

WHITING REQ# 79604 DATE 6-8-87

BY MJM PAGE 3-14 OF 27

*WPP 82587* ORIGINAL GIRDER



$$A = 1.25(17) + 1.25(21) + 2(.3125)(70.5) + 2(5.75) = 103.1 \text{ in}^2$$

## GIVEN DATA

1.	<u>72.25</u>	= DIMENSION A (IN.) -----	72.2500
2.	<u>21.</u>	= DIMENSION B (IN.) -----	21.0000
3.	<u>1.25</u>	= DIMENSION C (IN.) -----	1.2500
4.	<u>70.5</u>	= DIMENSION D (IN.) -----	70.5000
5.	<u>.3125</u>	= DIMENSION E (IN.) -----	0.3125
6.	<u>17.</u>	= DIMENSION F (IN.) -----	17.0000
7.	<u>1.25</u>	= DIMENSION G (IN.) -----	1.2500
8.	<u>.5</u>	= DIMENSION H (IN.) -----	0.5000
9.	<u>1.68</u>	= DIMENSION J (IN.) ( $\bar{Y}$ OF ANGLE) -----	1.6800
10.	<u>1.68</u>	= DIMENSION K (IN.) ( $\bar{X}$ OF ANGLE) -----	1.6800
11.	<u>5.75</u>	= AREA OF ANGLE (IN. <sup>2</sup> ) -----	5.7500
12.	<u>19.9</u>	= MOMENT OF INERTIA OF ANGLE (IN. <sup>4</sup> ) (VERT.) ---	19.9000
13.	<u>19.9</u>	= MOMENT OF INERTIA OF ANGLE (IN. <sup>4</sup> ) (HORIZ) ---	19.9000

## COMPUTED DATA

I <sub>y-y</sub> (IN. <sup>4</sup> ) (MOMENT OF INERTIA OF SECTION) -----	6,084.3136
$\bar{Y}$ (IN.) -----	38.2003
I <sub>x-x</sub> (IN. <sup>4</sup> ) (MOMENT OF INERTIA OF SECTION) -----	8,877.0612

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WHITING REQ. 79604 DATE 6-8-37  
 BY MJM PAGE 3-15 OF 24  
~~0021~~ 8.15.87

EFFECTIVE TORSIONAL PROPERTIES AT "CC" (REF 4 p 293 TABLE 20 CASE 16)

$$J = \frac{2 (.3125)(1.25) (17.3125)^2 (70)^2}{71.25(1.25) + 17.625(.3125) - 1.25^2 - .3125^2} = 12350 \text{ in}^4$$

EFFECTIVE PROPERTIES IN TRANSITION (REF 1 pp 4.44.2-4.44.3)

$$A = (110.6 + \sqrt{110.6(103.1)} + 103.1) / 3 = 106.8 \text{ in}^2$$

$$I_{xx} = \frac{158500 + \sqrt{(158500)^2(89880)} + \sqrt{(158500)(89880)} + \sqrt{(158500)(89880)^3} + 89880}{5}$$

$$= 121800 \text{ in}^4$$

$$I_{yy} = \frac{6285 + \sqrt{(6285)^2(6084)} + \sqrt{(6285)(6084)} + \sqrt{(6285)(6084)^3} + 6084}{5}$$

$$= 6184 \text{ in}^4$$

$$J = \frac{16880 + \sqrt{(16880)^2(12350)} + \sqrt{(16880)(12350)} + \sqrt{(16880)(12350)^3} + 12350}{5}$$

$$= 14530 \text{ in}^4$$

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# SECTIONAL PROPERTIES

(BUILT UP OF ROLLED &  
RECTANGULAR SECTIONS)

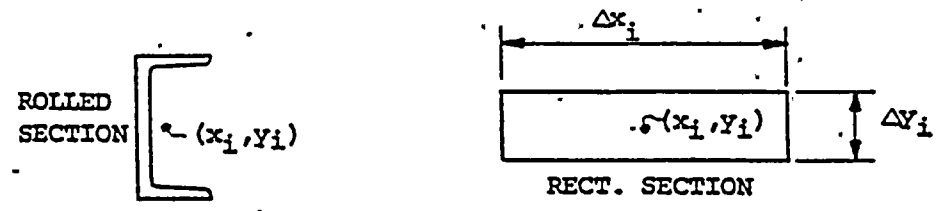
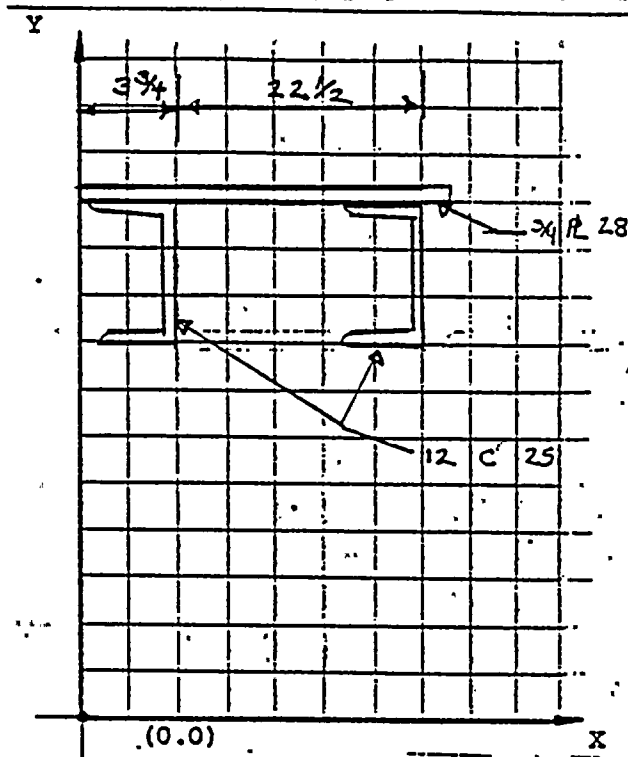
PROGRAM 116 PROGRAM ID 1-A-2-5(044)

WHITING REQ# 79604 DATE 6-8-37

BY MJM PAGE 3-16 OF 24

0.0018.25.87  
GIRDER END CONNECTION  
THRU MID

ELEMENT NO. (i)	ELEMENT PROPERTIES/ DIMENSIONS			ELEMENT CENTROID	
	$A_i$	$Ix_i$	$Iy_i$	$x_i$	$y_i$
ROLLED (1)					
1	7.35	144.	4.47	3.076	30.
2	7.35	144.	4.47	25.576	30.
	1000.				
RECT. (1000+i)	$\Delta x_i$	$\Delta y_i$	$x_i$	$y_i$	
1003	28.	.75	14.	36.375	
	-1				



All rolled sections must be entered before rectangular sections. To enter rectangular sections, end rolled sections with  $A_i = 1000$ .

In order to execute program enter a negative value for ' $A_i$ ' or ' $\Delta x_i$ '.

## COMPUTED DATA

- 14.1342 —  $\bar{x}$  - Distance from the 'y' axis to the centroid of the section.
- 33.7500 —  $\bar{y}$  - Distance from the 'x' axis to the centroid of the section.
- 35.7000 — A - Area of the section.
- 640.4062 —  $Ix$  - Moment of inertia about the section's neutral axis which is parallel to the 'x' axis.
- 3242.3277 —  $Iy$  - Moment of inertia about the section's neutral axis which is parallel to the 'y' axis.

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# SECTIONAL PROPERTIES

PROGRAM 116 PROGRAM ID 1-A-2-5(044)

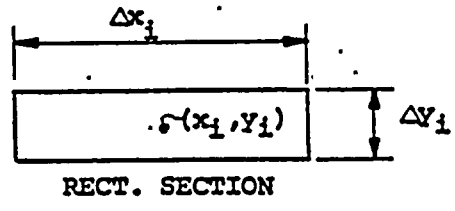
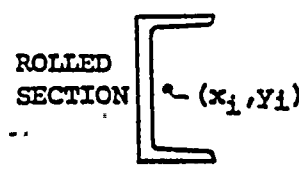
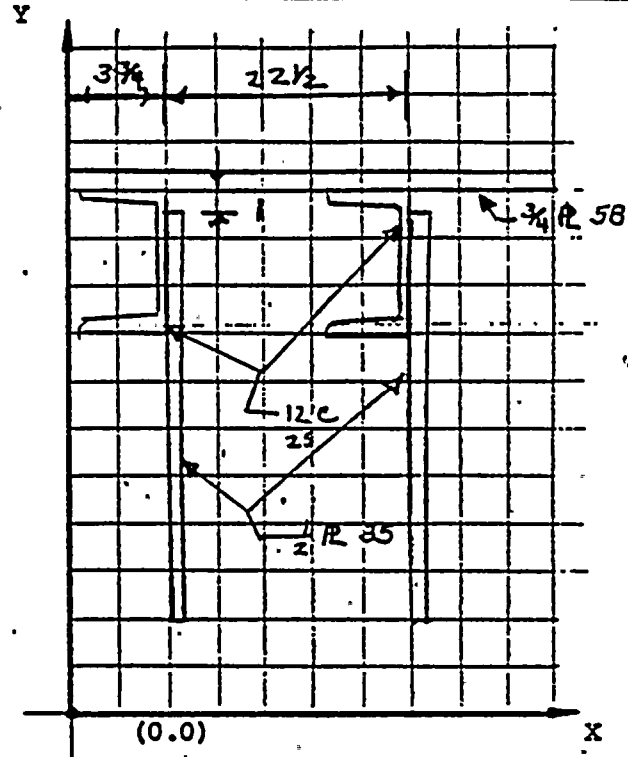
(BUILT UP OF ROLLED &  
RECTANGULAR SECTIONS)

WHITING REQ# 79604 DATE 6-8-57

BY MJM PAGE 3-17 OF 24

82587  
GIRDER END CONNECTION  
AT ENDS

ELEMENT NO. (i)	ELEMENT PROPERTIES/ DIMENSIONS			ELEMENT CENTROID	
	$A_i$	$I_{x_i}$	$I_{y_i}$	$x_i$	$y_i$
ROLLED (i)					
1	7.35	144.	4.47	3.076	30.
2	7.35	144.	4.47	25.576	30.
	1000.				
RECT. (1000+i)	$\Delta x_i$	$\Delta y_i$	$x_i$	$y_i$	
1003	.58	.75	29.	36.375	
1004	.5	35.	4.	17.5	
1005	.5	35.	26.5	17.5	
	-1.				



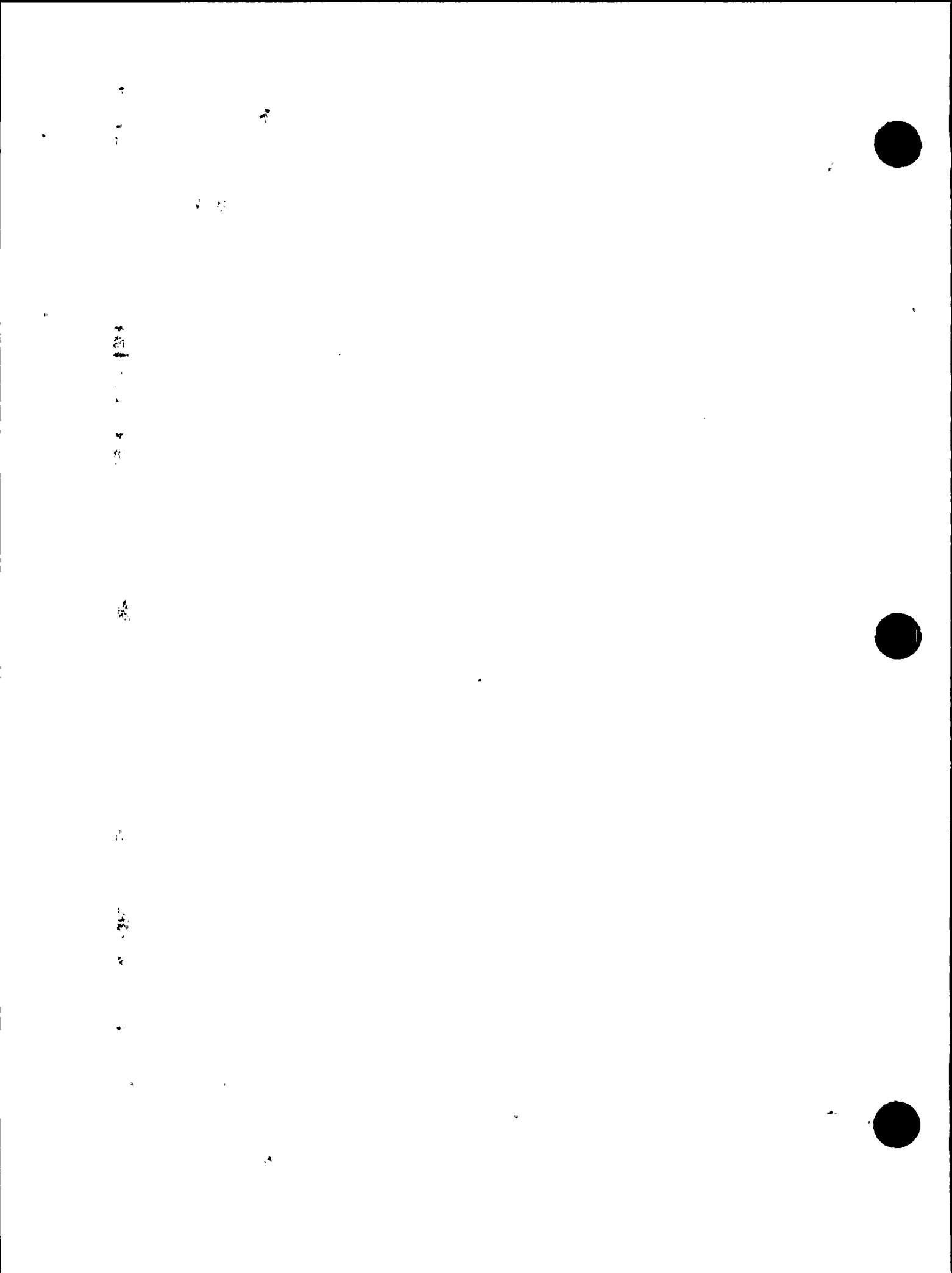
All rolled sections must be entered before rectangular sections. To enter rectangular sections, end rolled sections with  $A_i = 1000$ .

In order to execute program enter a negative value for ' $A_i$ ' or ' $\Delta x_i$ '.

## COMPUTED DATA

- 21.5219 ———  $\bar{x}$  - Distance from the 'y' axis to the centroid of the section.
- 28.2912 ———  $\bar{y}$  - Distance from the 'x' axis to the centroid of the section.
- 93.2000 ——— A - Area of the section.
- 10824.2506 ———  $I_x$  - Moment of inertia about the section's neutral axis which is parallel to the 'x' axis.
- 23064.8974 ———  $I_y$  - Moment of inertia about the section's neutral axis which is parallel to the 'y' axis.

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WHITING REQN. 79604 DATE 6-9-37  
 BY MJM PAGE 3-13 OF 24  
 82587

EFFECTIVE PROPERTIES  
 GIRDER END CONNECTION  
 AT ENDS

(REF. 1, pp 4.44.2-4.44.3)

$$A = (33.75 + \sqrt{33.75(93.20)} + 93.20) / 3 = 61.01 \text{ in}^2$$

$$I_{xx} = \frac{(640.4 + \sqrt{(640.4)^2(10820)} + \sqrt{(640.4)(10820)} + \sqrt{(640.4)(10820)^2 + 10820})}{5}$$

$$= 4146 \text{ in}^4$$

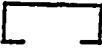
$$I_{yy} = \frac{(3242 + \sqrt{(3242)(23060)} + \sqrt{(3242)(23060)} + \sqrt{(3242)(23060)^2 + 23060})}{5}$$

$$= 10870 \text{ in}^4$$

EFFECTIVE TORSIONAL PROPERTIES

$$J = 2(.541) + 14(.375)^3 \left[ \frac{16}{3} - 3.36 \frac{.375}{14} \left( 1 - \frac{.375^4}{12(.4)^4} \right) \right]$$

$$= 4.953 \text{ in}^4 \quad (\text{REF 4, p 290, TABLE 20, CASE 4})$$

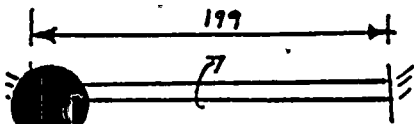
  
 (REF 4, p 300, table 21,  
 case 2)

$$e = 12 \frac{3(22)^2(12) + 6(22)^2(3) - 8(3)^3}{22^3 + 6(22)^2(12) + 6(22)^2(3) + 8(3)^3 - 12(22)(3)^2} = 5.976 \text{ in}$$

$$C_w = .625 \left[ \frac{(22)^2(12)^2}{2} \left( 3 + \frac{12}{3} - 5.976 - \frac{2(5.976)(3)}{12} + \frac{2(3)^2}{22} \right) \right. \\ \left. + \frac{(22)^2(5.976)^2}{2} \left( 12 + 3 + \frac{22}{6} - \frac{2(3)^2}{22} \right) + \frac{2(3)^3}{3} (12 + 5.976)^2 \right]$$

$$= 75090 \text{ in}^6$$

$$\theta = \left( \frac{4.953 (162 \times 10^6)}{75090 (29 \times 10^6)} \right)^{1/2} = .005047 \text{ in}^{-1}$$



$$J_{\text{eff}} = 4.953 \frac{\frac{1}{2} (199) (.005047)}{\left[ \frac{.005047(199)}{4} - \text{Tanh} \left[ \frac{.005047(199)}{4} \right] \right]}$$

$$= 483.3 \text{ in}^4$$

NO TWIST OR WARP  
 AT ENDS  
 (REF 4, p 298-316)

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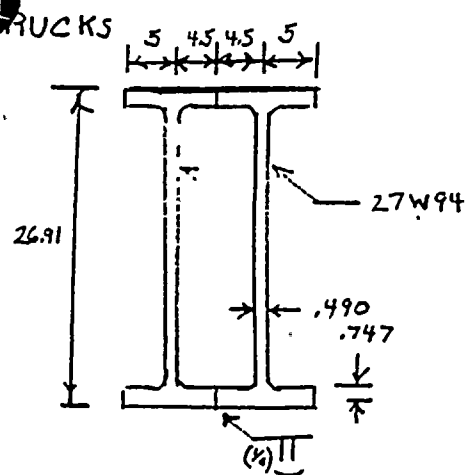
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WHITING REQN. 79604 DATE 6-8-37  
 BY MJM PAGE 3-12 OF 22  
~~8.25.87~~



$$A = 2(27.7) - 4(.5)(.747) = 53.91 \text{ in}^2$$

$$I_{xx} = 2(3270) - 4(.5)(.747)(13.08)^2$$

$$= 6284 \text{ in}^4$$

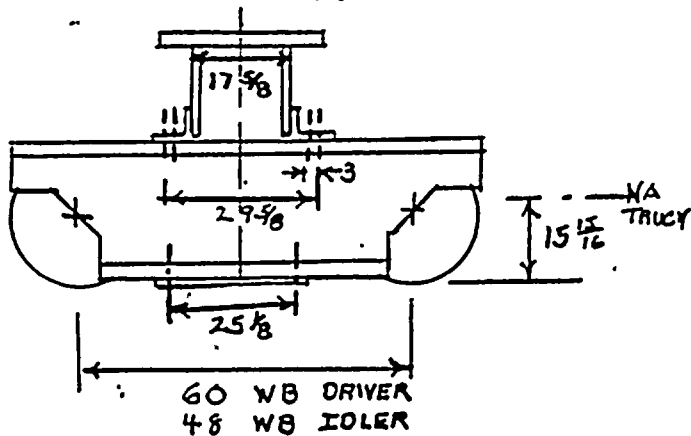
$$I_{yy} = 2 \left\{ [124 - 2(.5)(.747)(4.75)^2] + [27.7 - 2(.5)(.747)][4.5]^2 \right\}$$

$$= 1306 \text{ in}^4$$

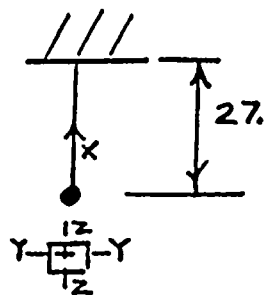
$$J_z = \frac{2(.490)(.747)(26.163)^2(9)^2}{26.91(.747) + 9.49(.49) - (.747)^2 - (.49)^2} = 169.4 \text{ in}^4$$

(REF 4, p 293, TABLE 20, CASE 16)

EFFECTIVE PROPERTIES OF BEAMS  
 TO SIMULATE TRUCKS

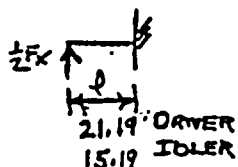


SIMULATE W/ BEAM



FOR FORCE IN X DIRECTION

(REF 3, p 104, TABLE III, CASE 1 & p 80)



$$\Delta X = \frac{\frac{1}{2} F_x l^3}{3 E I_{XTR}}$$

$$\Delta X = \frac{F_x l'}{A E}$$

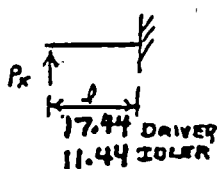
$$A = \frac{6 I_{XTR} l'}{l^3}$$

$$A_{DRIVER} = \frac{6 (6284) (27)}{(21.19)^3} = 107.0 \text{ in}^2$$

$$A_{IDLER} = \frac{6 (6284) (27)}{(15.19)^3} = 290.5 \text{ in}^2$$

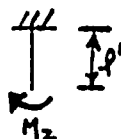
FOR MOMENT ABOUT Z AXIS

(REF 3, p 104, TABLE III, CASE 1 & 9)



$$P_x = \frac{M_z}{(WB)}$$

$$\theta = \frac{P_x l^2}{2 E I_{XTR}}$$



$$\theta = \frac{M_z l'}{E I_x}$$

$$I_z = \frac{2 (WB) I_{XTR} l'}{l^2}$$

$$I_z \text{ DRIVER} = \frac{2 (60) (6284) (27)}{(17.44)^2} = 66940 \text{ in}^4$$

$$I_z \text{ IDLER} = \frac{2 (48) (6284) (27)}{(11.44)^2} = 124500 \text{ in}^4$$



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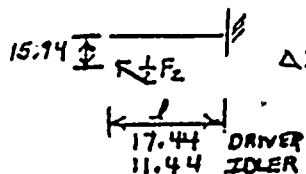
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FOR FORCE IN Z DIRECTION

(REF 3 P104 TABLE III CASE 18 P191)



$$\Delta Z = \frac{1}{2} F_z l^3 / 3EI_{yTR} + \frac{1}{2} F_z (15.59)^2 / J_{TG}$$

$$G = \frac{F}{2(1+.3)} = \frac{F}{2.6}$$



$$\Delta Z = \frac{F_z l^3}{3EI_y}$$

$$I_y = \frac{l^3}{3 \left( \frac{l^3}{6 I_{yTR}} + \frac{1.3(15.59)^2}{J_{TR}} \right)} = \frac{l^3}{\frac{l^3}{2 I_{ym}} + \frac{(991)l}{J_{TR}}}$$

$$I_{y \text{ DRIVER}} = \frac{(27)^3}{\frac{(17.44)^3}{2(1306)} + \frac{(991)(17.44)}{1694}} = 1609 \text{ in}^4$$

$$I_{y \text{ IDLER}} = \frac{(27)^3}{\frac{(11.44)^3}{2(1306)} + \frac{(991)(11.44)}{1694}} = 2709 \text{ in}^4$$

BECAUSE THE MODEL RESTRAINS THE BRIDGE ON ONE SIDE ONLY AND THE LOAD IS ASSUMED TO BE DISTRIBUTED TO THE RUNWAY ON A 2/3 1/3 RATIO TO ACCOUNT FOR FRICTIONAL RESISTANCE ON THE UNRESTRAINED SIDE:

$$I_{y \text{ DRIVER}} = \frac{1609}{2/3} = 2414 \text{ in}^4$$

$$I_{y \text{ IDLER}} = \frac{2709}{2/3} = 4064 \text{ in}^4$$



WHITING REQN. 79604 DATE 6-8-87  
 BY MJM PAGE 3-22 RI OF 24  
*OK 8.25.87*  
 REV MJM 9-15-87 *OK 9.15.87*

## TROLLEY

FRAME MODELED AS RIGID

MAIN HOIST ROPE

1 1/2 DIA PYTHON 10 F16 V - ALL STEEL - 26 STRAND - 362 WIRES  
 REEVED - 2 - 6 PART SYSTEMS - REDUNDANT (REF 8)

$$\text{AREA} = 12 \times \frac{(1\frac{1}{2})^2 \pi}{4} \times .629 = 13.34 \text{ in}^2$$

MODULUS OF ELASTICITY 14,000,000 PSI

## BRIDGE SPRING

THE BRIDGE SPRING IS USED AS A RIGID LINK FOR TRANSMITTING REACTIONS IN THE GLOBAL X DIRECTION (PARALLEL WITH THE GIRDERS). THE SPRING RATE IS SELECTED TO PROVIDE A NATURAL FREQUENCY IN THE HIGH FREQUENCY (STIFF) PORTION OF THE EXCITATION SPECTRUM. II

$$K = \frac{W}{g} (2\pi f)^2$$

$$= \frac{284000}{386.4} (2\pi(33))^2 = 31600000 \text{ lb/in}$$

WHERE

- K = SPRING RATE, lb/in  
 W = TOTAL WEIGHT OF CRANE, lb  
 (WITHOUT, LIFTED LOAD)  
 g = ACCELERATION OF GRAVITY, in/sec<sup>2</sup>  
 f = NATURAL FREQUENCY, Hz

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WHITING REQ. 79604 DATE 6-8-37  
 BY MJM PAGE 3-23 OF 24  
*0108 8.25.87*

MASS ELEMENTS

DESCRIPTION	NODE	WEIGHT LB	MASS LB SEC <sup>2</sup> /IN
TROLLEY LESS BLOCK	407	124000	320.9
BLOCK ONLY	406	20000	51.76
BLOCK W/ 50 T LIFTED LOAD	406	120000	310.6
BRIDGE DRIVE ASSEMBLY	502	4000	10.35
SINGLE REDUCTION UNITS	501,503	2000	5.18
DRIVE TRUCKS	103,203	5000	12.94
	101,201	2000	5.18
ISLER TRUCKS	104,204	4500	11.65
	102,202	2000	5.18

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WHITING REQN. 79604 DATE 6-8-87  
 BY MJM PAGE 3-24 OF 24  
*RJA 8.25.87*

DISTRIBUTED MASS  
 ON BEAM ELEMENTS

GIRDERS	DISTRIBUTED WEIGHT (LB/FT)	
	GIRDER A	GIRDER B
GIRDER INCLUDES STIFFENERS	470	470
RAIL	60	60
WALK	60	-
SQ SHAFT & CPLGS	20	-
CTRLS & SUPPT	80	-
BRIDGE CONDUCTOR	-	40
MISC (Incl RACK ON B)	<u>10</u>	<u>30</u>
	700	600

MASS DENSITY

GIRDER A

$$\frac{700}{12 \times 110.6 \times 386.4} = .001365 \text{ lb sec}^2/\text{in}^4$$

GIRDER B

$$\frac{600}{12 \times 110.6 \times 386.4} = .001170 \text{ lb sec}^2/\text{in}^4$$

END CONNECTIONS

DISTRIBUTED WEIGHT  
 180 lb/ft

MASS DENSITY

$$\frac{180}{12 \times 35.70 \times 386.4} = .001087 \text{ lb sec}^2/\text{in}^4$$



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WHITING REQN. 79604 DATE 8-26-87  
BY ASZ/MJM PAGE 4-1 OF 112  
*USA 8-27-87*

### SUPPLEMENTAL CALCULATIONS

This section summarizes the analysis of those components which were not directly analyzed by the finite element program utilizing the loadings that were generated by this program.

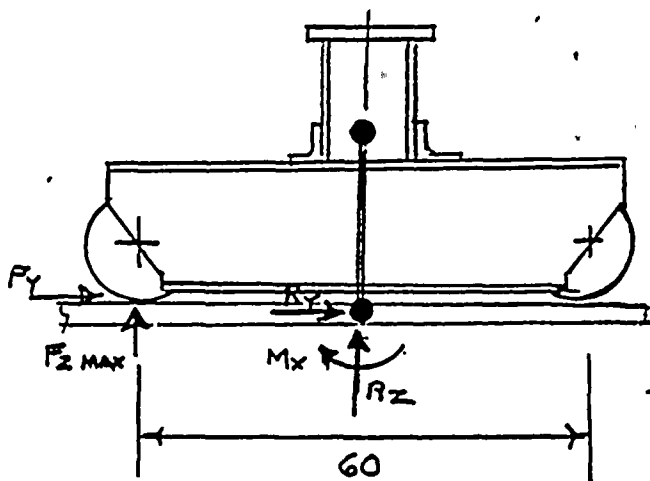
Page	Subject
4-2	Scale Factor
4-4	Bridge Truck Loads
4-7	Bridge Wheel Loads and Upkick
4-14	Bridge Wheels and Axles
4-16	Seismic Lugs
4-27	Bridge Trucks
4-37	Girder to Truck Connection
4-71	Girder to End Tie Connection
4-93	Girder Buckling Stability
4-104	Girder Welds
4-105	Girder End
4-108	Trolley Wheel Loads
4-111	Rope

For effects of seismic loads on trolley components see separate Structural Design Calculation Report.

WHITING REQN. 79604 DATE 6-8-37  
 BY MJM PAGE 4-2 OF 2  
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## SCALE FACTOR

BECAUSE SLIP WILL OCCUR AT THE RAIL WHEEL INTERFACE IF THE REACTION IN THE Y DIRECTION EXCEEDS THE MAXIMUM WHEEL LOAD IN THE Z DIRECTION TIMES THE COEFFICIENT OF FRICTION, THE ACCELERATION IN THE Y DIRECTION WILL BE LESS THAN PREDICTED BY A MODAL ANALYSIS. THE PRIMARY Y MODE MAY BE PROPORTIONED BY A SCALE FACTOR THAT ACCOUNTS FOR SLIDING AND THAT IS DERIVED AS FOLLOWS:



WHERE

$R_z$ ,  $R_y$  &  $M_x$   
 ARE MAXIMUM  
 REACTIONS FROM  
 FINITE ELEMENT  
 MODAL ANALYSIS

AND

$F_{z \text{ MAX}}$  IS MAX  
 Z REACTION AT  
 A DRIVE WHEEL  
 $F_y$  IS MAX Y  
 REACTION BY  
 FRICTION

SCF IS SCALE FACTOR

$$F_{z \text{ MAX}} = \frac{R_z}{2} + \frac{M_x}{60}$$

$$F_y = .25 F_{z \text{ MAX}} \quad (\text{REF 15, P 3-38})$$

$$\text{SCF} = \frac{F_y}{R_y}$$

OBSERVING THAT  $M_x$  IS DUE PRIMARILY  
 TO Y EXCITATIONS.

$$\text{SCF} = \frac{.25}{R_y} \left( \frac{R_z}{2} + \frac{\text{SCF}(M_x)}{60} \right)$$

$$\text{SCF} = \frac{R_z}{8 R_y \left( 1 - \frac{M_x}{240(R_y)} \right)}$$



WHITING REQ. 79604 DATE 6-17-37  
 BY MJM PAGE 4-3 OF 11  
 6758.2:27

TABLE 4-1  
SCALE FACTORS

TROLLEY	LOAD	OBE	SSE
MID	50T UP	.1096	.0744
	50T DN	.1420	.1078
1/4	50T UP	.1024	.0686
	50T DN	.1436	.1074
END	50T UP	.2260	.1356
	50T DN	.3365	.2385
MID	NO LOAD	.0801	.0554
1/4	NO LOAD	.0775	.0521
RHE	NO LOAD	.3696	.2156

FOR REACTIONS SEE TABLES B19 through B54

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WHITING REQN. 72504 DATE 3-7-87  
BY ASZ PAGE 4-4 OF 2  
MJM 8-24-87

## TRUCK LOADS

THE  $R_x$  AT 124 IS DIVIDED BY 2 FOR THE 2 TRUCKS ON THE HELD SIDE AND MULTIPLIED BY  $\frac{2}{3}$  TO ACCOUNT FOR FRICTIONAL RESISTANCE AT THE UNRESTRAINED WHEELS. THE MINIMUM TRUCK LOAD IS THE NEGATIVE OF THE MAXIMUM CONSIDERING COMPLETE REVERSAL.

$$F_x = \frac{2}{3} \left( \frac{R_x}{2} \right) = \frac{R_x}{3}$$

ALL OTHER TRUCK LOADS ARE SUMMARIZED DIRECTLY FROM TABLES B19 TO B54 AFTER THE APPLICATION OF THE SCALE FACTOR AS PREVIOUSLY DESCRIBED.

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WHITING REQN. 7250 DATE 5-1-87  
 BY ASZ PAGE 4-5 OF 12  
 MJM 8-24-87

TABLE 4-2

## TRUCK LOADS (OBE - SCALED)

	TROLLEY	LOAD	SUM (MAX)				DIFFER (MIN.)			
			F <sub>x</sub>	F <sub>y</sub>	F <sub>z</sub>	M <sub>x</sub>	F <sub>x</sub> '	F <sub>y</sub> '	F <sub>z</sub> '	M <sub>x</sub> '
DRIVER (NODE 101,201)	MID	UP	11.2	26.5	141.7	1869	-11.2	-26.5	60.1	-1781
		DN	11.3	34.3	192.6	2338	-11.3	-34.3	9.3	-2250
		NO	10.9	19.4	90.9	1339	-10.9	-19.4	60.2	-1252
	1/4	UP	8.57	32.0	159.5	2200	-8.57	-32.0	54.3	-2113
		DN	11.8	44.1	242.8	3011	-11.8	-44.1	25.1	-2924
		NO	10.6	25.0	110.8	1730	-10.6	-25.0	46.5	-1647
	END	UP	9.17	39.5	165.8	2717	-9.17	-39.5	44.7	-2633
		DN	10.3	49.8	250.2	3267	-10.3	-49.8	31.9	-3184
		NO	9.16	43.9	116.4	2970	-9.16	-43.9	37.1	-2891
IDLER (NODE 102,202)	MID	UP	11.2	0	130.0	458.8	-11.2	0	55.1	-278.3
		DN	11.3	0	180.9	204.2	-11.3	0	4.55	-23.7
		NO	10.9	0	80.9	171.4	-10.9	0	53.9	7.5
	1/4	UP	8.57	0	145.6	382.8	-8.57	0	50.3	-203.0
		DN	11.8	0	227.6	374.4	-11.8	0	17.7	-194.6
		NO	10.6	0	97.6	365.7	-10.6	0	42.3	-188.3
	END	UP	9.17	0	152.8	892.1	-9.17	0	40.1	-710.2
		DN	10.3	0	237.3	933.4	-10.3	0	27.3	-751.5
		NO	9.16	0	105.0	940.9	-9.16	0	32.0	-772.2

ALL FORCES IN KIPS, MOMENTS IN IN.KIP IN GLOBAL COORDINATE  
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WHITING REQ. 79604 DATE 6-17-87  
 BY ASZ PAGE 4-6 OF 112  
 MJM 8-24-87

TABLE 4-3

## TRUCK LOADS (SSE - SCALED)

	TROLLEY	LOAD	SUM (MAX)				DIFFER (MIN.)			
			F <sub>x</sub>	F <sub>y</sub>	F <sub>z</sub>	M <sub>x</sub>	F' <sub>x</sub>	F' <sub>y</sub>	F' <sub>z</sub>	M' <sub>x</sub>
DRIVER (NODE 101,201)	MID	UP	21.3	33.8	172.7	2436	-21.3	-33.8	30.2	-2348
		DN	21.5	48.9	271.9	3313	-21.5	-48.9	-67.8	-3224
		NO	20.9	25.2	109.7	1728	-20.9	-25.2	42.2	-1641
	1/4	UP	17.2	42.2	194.2	2894	-17.2	-42.2	38.2	-2808
		DN	22.2	63.0	338.0	4313	-22.2	-63.0	-69.5	-4227
		NO	20.4	33.9	129.1	2349	-20.4	-33.9	36.7	-2265
	END	UP	18.6	62.2	189.4	4456	-18.6	-62.2	35.1	-4373
		DN	19.8	77.2	339.4	5224	-19.8	-77.2	-42.6	-5141
		NO	18.3	78.8	127.4	5384	-18.3	-78.8	29.9	-5305
IDLER (NODE 102,202)	MID	UP	21.3	0	158.3	743.6	-21.3	0	27.6	-563.1
		DN	21.5	0	257.0	268.5	-21.5	0	-69.2	-88.0
		NO	20.9	0	97.0	224.9	-20.9	0	38.8	-46.0
	1/4	UP	17.2	0	176.5	614.9	-17.2	0	35.3	-435.1
		DN	22.2	0	319.4	605.6	-22.2	0	-74.1	-425.8
		NO	20.4	0	112.6	581.4	-20.4	0	33.5	-404.1
	END	UP	18.6	0	173.2	1521	-18.6	0	31.6	-1339
		DN	19.8	0	324.8	1573	-19.8	0	-50.4	-1391
		NO	18.3	0	114.0	1626	-18.3	0	25.6	-1457

ALL FORCES IN KIPS, MOMENTS IN IN.KIP IN GLOBAL COORDINATE  
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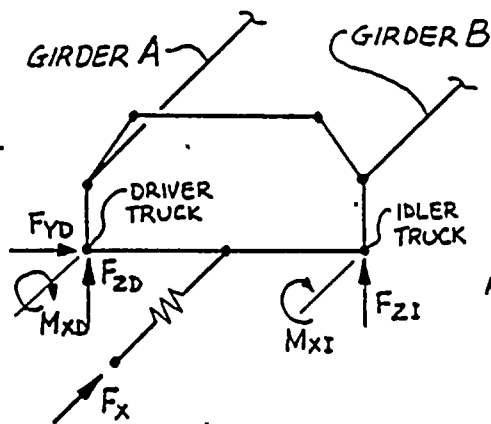
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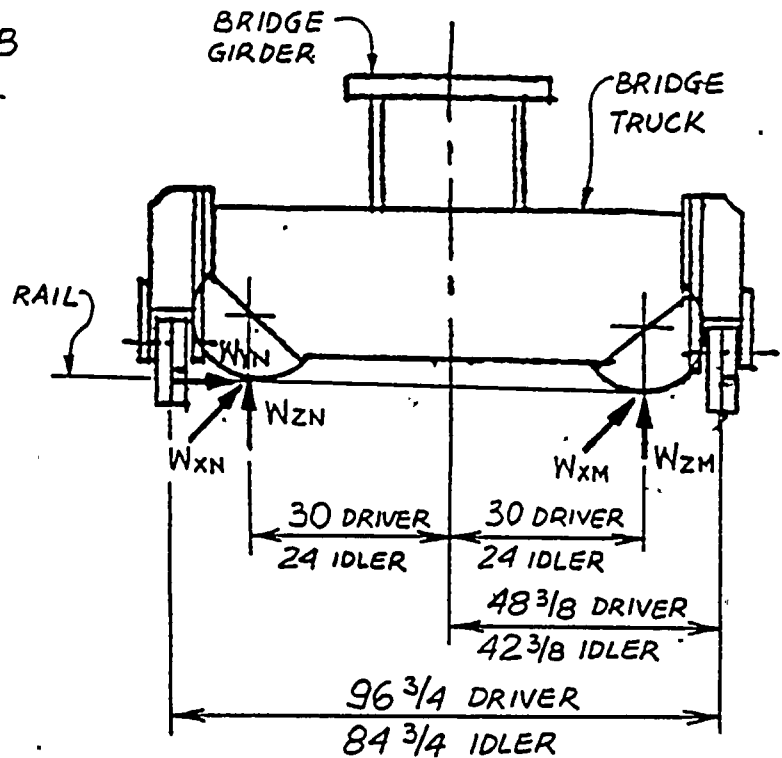
WHITING REQ. 79604 DATE 8-18-87  
 BY ASZ PAGE 4-7 OF 12  
 MJM 8-24-87

BRIDGE WHEEL LOADS AND UPKICK

THE WHEEL LOADS ARE DETERMINED BY APPLYING THE BRIDGE TRUCK REACTIONS, SHOWN IN SCALED TABLES B19 TO B54, TO THE TRUCKS IN THE FOLLOWING MANNER :



MODEL SCHEMATIC OF END OF CRANE



BRIDGE DRIVER TRUCK IS LOCATED ON GIRDER A AND IDLER TRUCK IS LOCATED ON GIRDER B.

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WHITING REQN. 79604 DATE 8-18-87  
 BY ASZ PAGE 4-9 OF 12  
 MJM 3-24-87

### FLANGING WHEEL LOADS.

$W_{XMAX} \rightarrow F_x$  (FROM SCALED TABLES B19 TO B54) IS DIVIDED BETWEEN OPPOSITE ENDS OF CRANE IN A  $\frac{2}{3}$  AND  $\frac{1}{3}$  MANNER. FOR THE FOUR WHEELS ON THE HELD SIDE  $F_x$  IS DIVIDED BY 4 AND MULTIPLIED BY  $\frac{2}{3}$  TO ACCOUNT FOR FRICTIONAL RESISTANCE AT THE UNRESTRAINED WHEELS.

$$W_{XMAX} = \frac{1}{4} \left( \frac{2}{3} \right) F_x = \frac{F_x}{6}$$

$W_{XMIN} \rightarrow$  THE MINIMUM WHEEL LOAD IS THE NEGATIVE OF THE MAXIMUM CONSIDERING COMPLETE REVERSAL

$$W_{XMIN} = - W_{XMAX}$$

### VERTICAL WHEEL LOADS AND UPKICK LOADS.

#### WHEEL LOADS.

$W_{ZMAX}, W_{ZMIN} \rightarrow$  IN DETERMINING WHEEL LOADS THE MAX. TRUCK REACTIONS

$F_z$  AND  $M_x$  WERE TAKEN FROM THE "SUM" COLUMN OF SCALED

TABLES B19 TO B54.

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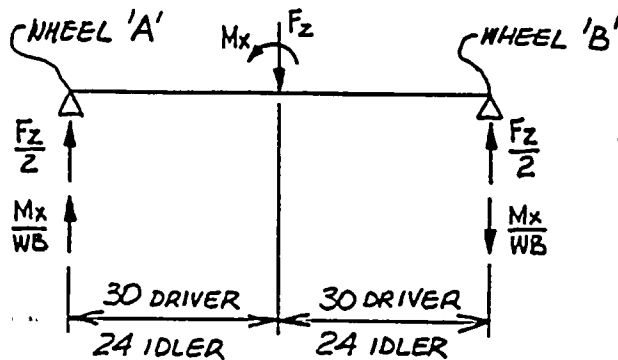




WHITING REQ. 79604 DATE 8-18-37  
 BY ASZ PAGE 4-9 OF 12  
 MJM 8-24-87

IF  $\frac{F_z}{2} \leq \frac{M_x}{WB}$  THEN  $W_{ZMIN} = 0$  AND UPKICK OCCURS.

∴ FROM UP-KICK CALCULATIONS FOR THIS CONDITION  $N_{ZMAX} = f_A F_z + \frac{M_x}{SP}$



$W_{ZMIN} = 0$  AND  
 $UPKICK = N_{ZMAX} - F_z$   
 (SEE PG. 4-9)  
 WHEEL BASE  $WB = 60$  IN. (DRIVER)  
 $= 48$  IN. (IDLER)

$$W_{ZMAX} = W_{ZA} = \frac{F_z}{2} + \frac{M_x}{WB} \quad W_{ZMIN} = W_{ZB} = \frac{F_z}{2} - \frac{M_x}{WB}$$

### UPKICK LOAD

$P_{UL}$ ,  $P_{UR}$  → IN DETERMINING UPKICK LOADS, THE LOAD  $F_z$  WAS TAKEN FROM THE "DIFFERENCE" COLUMN AND MOMENT  $M_x$  WAS TAKEN FROM THE "SUM" COLUMN OF SCALED TABLES B 19 TO B 54. ROPE UPKICK LOAD ( $R_U$ ) WAS TAKEN FROM TABLES B 62 AND B 64 FOR LOAD IN DOWN POSITION. FOR LOADS IN UP POSITION AND FOR THE NO-LOAD CONDITION ROPE UPKICK LOAD ( $R_U$ ) DOES NOT EXIST AND THEREFORE EQUALS ZERO.

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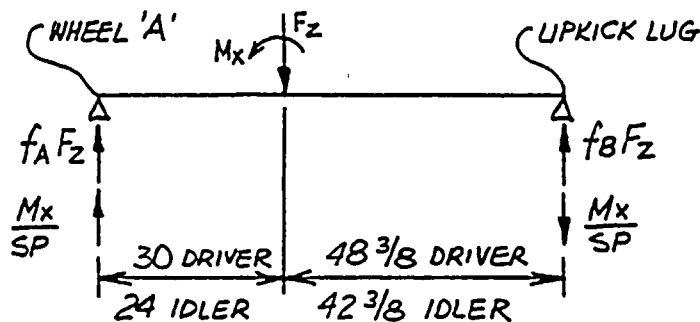
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102



WHITING REQ. 79604 DATE 8-18-87  
 BY ASZ PAGE 4-10 OF 12  
 MJM 8-24-87

$$IF \frac{1}{2} \left[ F_z - \frac{x_i}{BS} \frac{R_u}{2} \right] < \frac{M_x}{WB} \text{ THEN UPKICK OCCURS}$$



$$\begin{aligned} \text{SPAN SP} &= 78 \frac{3}{8} \text{ IN. (DRIVER)} \\ &= 66 \frac{3}{8} \text{ IN. (IDLER)} \end{aligned}$$

$$\text{LOAD FACTOR} = f$$

$$\underline{\underline{A}} \quad \underline{\underline{DRIVER}} \quad f_A = \frac{48.375}{78.375} = 0.62$$

$$\underline{\underline{A}} \quad \underline{\underline{IDLER}} \quad f_A = \frac{42.375}{66.375} = 0.64$$

$$\underline{\underline{B}} \quad \underline{\underline{DRIVER}} \quad f_B = \frac{30}{78.375} = 0.38$$

$$\underline{\underline{B}} \quad \underline{\underline{IDLER}} \quad f_B = \frac{24}{66.375} = 0.36$$

$$W_z = f_A (F_z) + \frac{M_x}{SP}$$

$$P_{ui} = -f_B \left( F_z i - \frac{x_i}{BS} \frac{R_u}{2} \right) + \frac{M_x i}{SP}$$

$i = L$  FOR BRIDGE LHE AND  $i = R$  FOR BRIDGE RHE

$F_{zL}$  AND  $M_{xL}$  WERE TAKEN AT NODES 201 FOR DRIVER AND 202 FOR IDLER FROM SCALED TABLES B19 TO B54.

$F_{zR}$  AND  $M_{zR}$  WERE TAKEN AT NODES 101 FOR DRIVER AND 102 FOR IDLER FROM SCALED TABLES B19 TO B54.

$X$  IS A DISTANCE BETWEEN BRIDGE END AND MAIN TROLLEY HOOK AND DEPENDS ON TROLLEY POSITION AND EQUALS:

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WHITING REQ. 79604 DATE 8-11-87  
 BY ASZ PAGE 4-11 OF 2  
 MJM 8-24-87

FOR MAIN TROLLEY AT MID.  $X_L = 466$  IN.  $X_R = 466$  IN.  
 QUARTER  $X_L = 233$  IN.  $X_R = 699$  IN.  
 END  $X_L = 123.25$  IN.  $X_R = 808.75$  IN.

BRIDGE SPAN  $BS = 932$  IN.

SINCE THE LINEAR COMPUTER ANALYSIS SHOWS THE HOIST ROPE GOING IN COMPRESSION (SLACK ROPE CONDITION) WHEN THE LOAD IS IN THE DOWN POSITION, THE UPKICK LOADS WERE DETERMINED BY SUBTRACTING THE RELATIVE PROPORTION OF THE ROPE COMPRESSIVE LOAD  $[R_u]$  (WHICH CANNOT EXIST) FROM THE VERTICAL REACTIONS.

TANGENTIAL WHEEL LOADS (BRAKE WHEEL ON DRIVER TRUCK ONLY).

$W_{YMAX}$  → THE MAXIMUM WHEEL LOAD IN THE Y DIRECTION IS TAKEN TO BE LIMITED BY THE COEFFICIENT OF FRICTION OF 0.25 AND THE MAXIMUM WHEEL LOAD ( $W_{ZMAX}$ ).

$$W_{YMAX} = 0.25 W_{ZMAX}$$

THE MINIMUM WHEEL LOAD IS THE NEGATIVE OF THE MAXIMUM CONSIDERING COMPLETE REVERSAL.

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WHITING REQ. 79604 DATE 8-18-87  
 BY ASZ PAGE 4-12 OF 12  
 MJM 8-24-87

TABLE 4-4

## CRANE WHEEL LOADS

OBE SCALED

	TROLLEY	LOAD	W <sub>X</sub> MAX	W <sub>Y</sub> MAX	MAX. W <sub>Z</sub>		P <sub>UL</sub> 201, 202	P <sub>UR</sub> 101, 102	TABLE USED
					W <sub>A</sub> (MAX)	W <sub>B</sub> **			
DRIVER 101, 201	MID	UP	5.60	25.5	102.0	39.7	-	-	B20
		DN	5.64	33.8	135.3	57.3	12.0	13.3	B24
	1/4	UP	4.28	29.1	116.4	43.1	-	-	B28
		DN	5.88	42.9	171.6	71.2	1.3	12.6	B32
	END	UP	4.59	32.0	128.2	37.6	0.3	-	B36
		DN	5.13	44.9	179.6	70.6	3.8	8.5	B40
	MID	NO	5.46	16.9	67.8	23.1	-	-	B44
	1/4	NO	5.31	21.1	84.2	26.6	-	-	B48
	END	NO	4.58	26.9	107.7	8.7	7.0	0.5	B52
	IDLER 102, 202	MID	UP	5.60	-	74.6	55.4	-	-
DN			5.64	-	94.7	86.2	-	-	B24
1/4		UP	4.28	-	80.8	64.8	-	-	B28
		DN	5.88	-	121.6	106.0	-	-	B32
END		UP	4.59	-	95.0	57.8	-	-	B36
		DN	5.13	-	138.1	99.2	-	-	B40
MID		NO	5.46	-	44.0	36.9	-	-	B44
1/4		NO	5.31	-	56.4	41.2	-	-	B48
END		NO	4.58	-	71.7	33.3	2.7	-	B52

ALL FORCES IN KIPS IN GLOBAL COORDINATE SYSTEM.

\* INDICATES UPKICK LOAD AT UPKICK LUG FOR THE STATIC PLUS DYNAMIC CONDITION.

\*\* W<sub>B</sub> IS LOAD ON OTHER WHEEL OF TRUCK WHEN W<sub>A</sub> IS MAX.

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WHITING REQ. 79604 DATE 8-18-87  
 BY ASZ PAGE 4-13 OF 12  
 MJM 8-24-87

TABLE 4-5

CRANE WHEEL LOADS

SSE SCALED

	TROLLEY	LOAD	W <sub>X</sub> MAX	W <sub>Y</sub> MAX	MAX. W <sub>Z</sub>		P <sub>UL</sub> 201,202	P <sub>UR</sub> 101,102	TABLE USED
					W <sub>A</sub> (MAX)	W <sub>B</sub> **			
DRIVER 101,201	MID	UP	10.6	31.7	127.0	45.7	15.8	19.2	B22
		DN	10.8	47.8	191.2	80.7	32.5	35.4	B26
	1/4	UP	8.6	36.3	145.3	48.9	2.3	8.7	B30
		DN	11.1	60.2	240.9	97.1	12.5	36.3	B34
	END	UP	9.3	42.2	169.0	20.4	17.1	16.0	B38
		DN	9.9	64.2	256.8	82.6	19.3	36.6	B42
	MID	NO	10.4	20.9	83.7	26.0	4.3	5.2	B46
	1/4	NO	10.2	25.9	103.7	25.4	0.1	5.5	B50
	END	NO	9.2	38.4	153.4	-(26.0*)	26.4	35.5	B54
	IDLER 102,202	MID	UP	10.6	-	94.6	63.7	-	0.6
DN			10.8	-	134.1	122.9	-	-	B26
1/4		UP	8.6	-	101.1	75.4	-	-	B30
		DN	11.1	-	172.3	147.1	-	-	B34
END		UP	9.3	-	118.3	54.9	9.4	-	B38
		DN	9.9	-	195.2	129.6	10.6	-	B42
MID		NO	10.4	-	53.2	43.8	-	-	B46
1/4		NO	10.2	-	68.4	44.2	-	-	B50
END		NO	9.2	-	88.6	25.4	15.3	-	B54

ALL FORCES IN KIPS IN GLOBAL COORDINATE SYSTEM.

\* INDICATES UPKICK LOAD AT UPKICK LUG FOR THE STATIC PLUS DYNAMIC CONDITION.

\*\* W<sub>B</sub> IS LOAD ON OTHER WHEEL OF TRUCK WHEN W<sub>A</sub> IS MAX.

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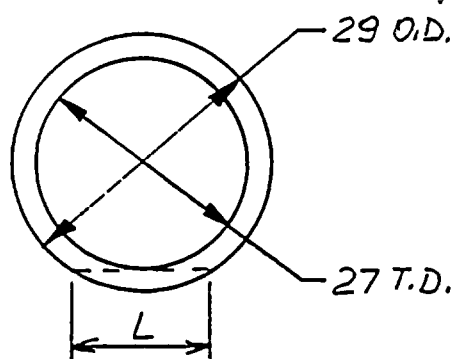
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WHITING REQ. 73504 DATE 5-3-37  
 BY ASZ PAGE 4-14 OF 12  
 MJM8-24-87

## BRIDGE WHEEL

### FLANGE SHEAR STRESS



MTRL: ROLLED STEEL

$$\sigma_{YMIN} = 53 \text{ KSI}$$

$$\text{OBE } \tau_{ALL} = 0.6 \frac{\sigma_{YMIN}}{1.5} = 21.2 \text{ KSI}$$

$$\text{SSE } \tau_{ALL} = 0.6 \frac{\sigma_{YMIN}}{1.1} = 28.9 \text{ KSI}$$

$$F_x = 5.88 \text{ KIP (MAX. FOR OBE FROM TABLE 4-4)}$$

$$= 11.1 \text{ KIP (MAX. FOR SSE FROM TABLE 4-5)}$$

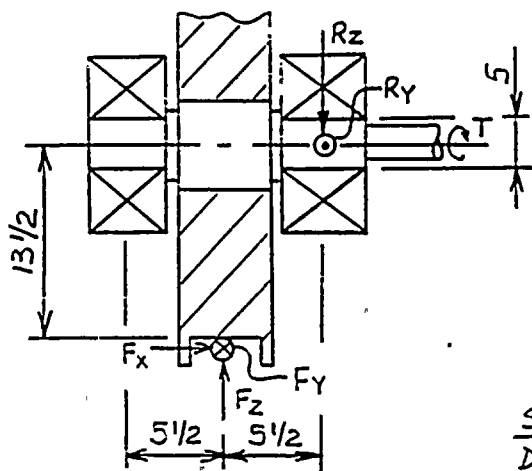
$$L = 2 \left[ \left( \frac{29}{2} \right)^2 - \left( \frac{27}{2} \right)^2 \right]^{1/2} = 10.6 \text{ IN.}$$

ASSUME ONLY HALF EFFECTIVE IN SHEAR

$$\text{OBE } \tau = \frac{5.88}{0.5 \times 10.6 \times 0.875} = 1.3 \text{ KSI}$$

$$\text{SSE } \tau = \frac{11.1}{0.5 \times 10.6 \times 0.875} = 2.4 \text{ KSI}$$

## BRIDGE AXLE



MTRL: AISI-1144 HT, 220-260 BHN

$$\sigma_{YMIN} = 60 \text{ KSI}$$

$$\text{OBE } \tau_{ALL} = 0.6 \frac{\sigma_{YMIN}}{1.5} = 24 \text{ KSI}$$

$$\text{SSE } \tau_{ALL} = 0.6 \frac{\sigma_{YMIN}}{1.1} = 32.7 \text{ KSI}$$

$$\frac{\text{SPAN}}{\text{DEPTH}} = \frac{11}{5} = 2.2 < 3$$

∴ MODE OF FAILURE IS SHEAR FOR SEISMIC LOADS

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WHITING REQ. 7250- DATE 6-3-37  
 BY ASZ PAGE 4-5 OF 12  
 MJM 8-24-87

FROM TABLE 4-4 FOR DBE MAX. LOADS ARE:

$$\text{DRIVER } W_{X\text{MAX}} = 5.88 \text{ KIP} \quad W_{Y\text{MAX}} = 44.9 \text{ KIP} \quad W_{Z\text{MAX}} = 179.6 \text{ KIP}$$

$$\text{IDLER } W_{X\text{MAX}} = 5.88 \text{ KIP} \quad W_{Y\text{MAX}} = 0 \quad W_{Z\text{MAX}} = 138.1 \text{ KIP}$$

FROM TABLE 4-5 FOR SSE MAX. LOADS ARE:

$$\text{DRIVER } W_{X\text{MAX}} = 11.1 \text{ KIP} \quad W_{Y\text{MAX}} = 64.2 \text{ KIP} \quad W_{Z\text{MAX}} = 256.8 \text{ KIP}$$

$$\text{IDLER } W_{X\text{MAX}} = 11.1 \text{ KIP} \quad W_{Y\text{MAX}} = 0 \quad W_{Z\text{MAX}} = 195.2 \text{ KIP}$$

OBE

$$\text{DRIVER } R_{RD} = \sqrt{\left(\frac{W_{Y\text{MAX}}}{2}\right)^2 + \left(\frac{W_{Z\text{MAX}}}{2} + \frac{W_{X\text{MAX}} \cdot \text{TD}}{\text{SPAN}}\right)^2} = \sqrt{\left(\frac{44.9}{2}\right)^2 + \left(\frac{179.6}{2} + \frac{5.88 \times 13.5}{11}\right)^2}$$

$$= 99.6 \text{ KIP}$$

$$\text{IDLER } R_{RI} = \frac{138.1}{2} + \frac{5.88 \times 13.5}{11} = 76.3 \text{ KIP}$$

$$\tau_{\text{MAX}} = \frac{4}{3} \frac{R_{R\text{MAX}}}{A} = \frac{4}{3} \frac{99.6}{\frac{\pi \times 5^2}{4}} = 6.8 \text{ KSI}$$

SSE

$$\text{DRIVER } R_{RD} = \sqrt{\left(\frac{64.2}{2}\right)^2 + \left(\frac{256.8}{2} + \frac{11.1 \times 13.5}{11}\right)^2} = 145.6 \text{ KIP}$$

$$\text{IDLER } R_{RI} = \frac{195.2}{2} + \frac{11.1 \times 13.5}{11} = 111.2 \text{ KIP}$$

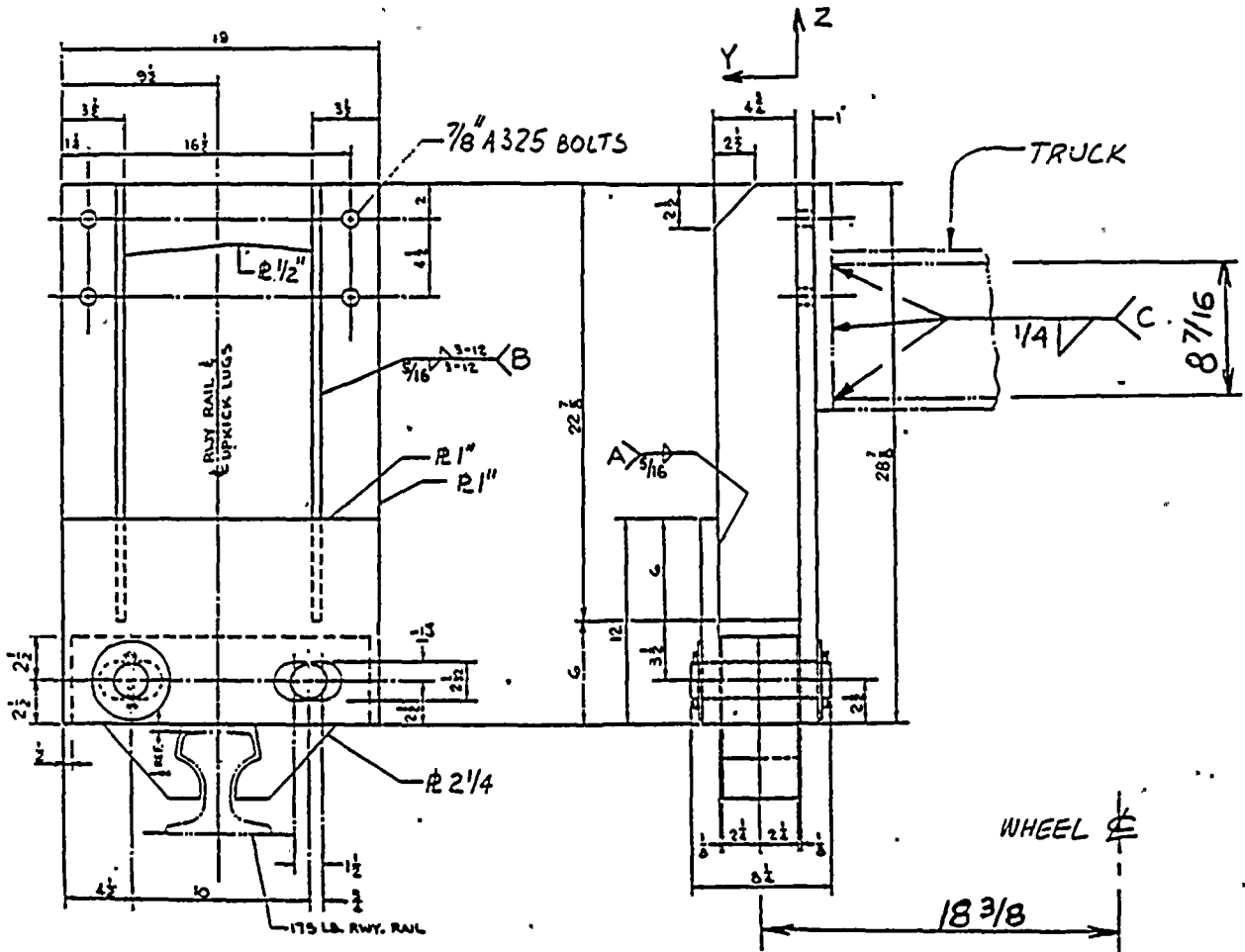
$$\tau_{\text{MAX}} = \frac{4}{3} \frac{145.6}{\frac{\pi \times 5^2}{4}} = 9.9 \text{ KSI}$$

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WHITING REQN. 79524 DATE 2-13-37  
 BY ASZ PAGE 4-16 OF 2  
 NJM 0-24-87

BRIDGE TRUCK SEISMIC LUGS.



MTRL.: ASTM-A36

$G_{MIN} = 36 \text{ KSI}$

OBE  $G_{ALL} = \frac{G_{MIN}}{1.5} = 24 \text{ KSI}$

$\tau_{ALL} = 0.6 G_{ALL} = 14.4 \text{ KSI}$

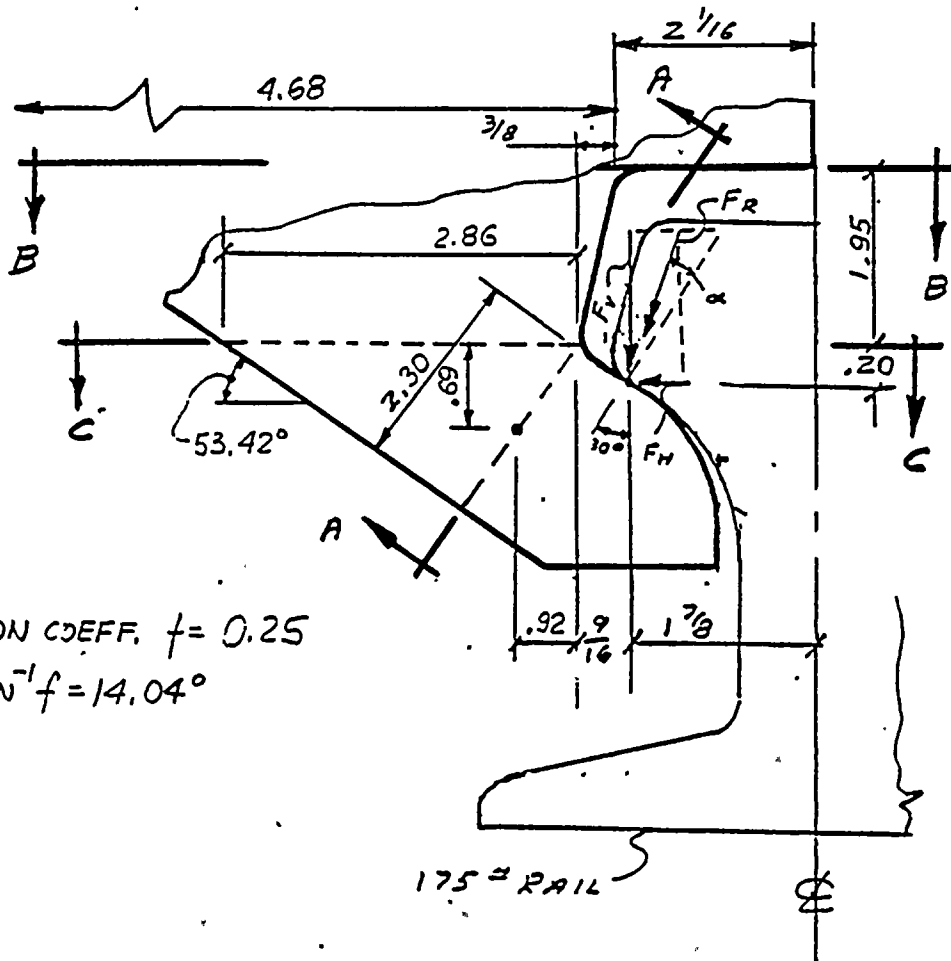
SSE  $G_{ALL} = \frac{G_{MIN}}{1.1} = 32.7 \text{ KSI}$

$\tau_{ALL} = 0.6 G_{ALL} = 19.6 \text{ KSI}$





WHITING REQ. 79604 DATE 3-13-87  
 BY ASZ PAGE 4-17 OF 22  
 MJM 8-24-87



FRICITION COEFF.  $f = 0.25$   
 $\alpha = \tan^{-1} f = 14.04^\circ$

MAX. WHEEL UPKICK LOAD  $P_{MAX} = P_{UR}$  FROM TABLE 4-4 TO 4-5

OBE  $P_{MAX} = 13.3$  KIP

SSE  $P_{MAX} = 36.6$  KIP

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WHITING REQ. 79504 DATE 2-15-87  
 BY ASZ PAGE 4-8 OF 12  
 MJM 8-24-87

LOAD ON EACH OF THE CONTOURED LUGS :

$$F_V = \frac{P_{MAX}}{2} \quad F_H = F_V \tan(30 - \alpha) = 0.5 P_{MAX} \tan(30 - 14.04)$$

$$= 0.143 P_{MAX}$$

RESULTANT  $F_R = \sqrt{F_V^2 + F_H^2} = \sqrt{(0.5 P_{MAX})^2 + (0.143 P_{MAX})^2}$

$$= 0.52 P_{MAX}$$

OBE  $F_V = 0.5 \times 13.3 = 6.7 \text{ KIP}$   $F_H = 0.143 \times 13.3 = 1.9 \text{ KIP}$

$$F_R = 0.52 \times 13.3 = 6.9 \text{ KIP}$$

SSE  $F_V = 0.5 \times 36.6 = 18.3 \text{ KIP}$   $F_H = 0.143 \times 36.6 = 5.2 \text{ KIP}$

$$F_R = 0.52 \times 36.6 = 19.0 \text{ KIP}$$

SECTION A-A (SHEAR)

$$\tau_A = \frac{F_R \cos(53.42^\circ - 30^\circ + 14.04)}{A_A}$$

CBE  $\tau_A = \frac{6.9 \cos 37.46^\circ}{2.25 \times 2.3} = 1.1 \text{ KSI}$

SSE  $\tau_A = \frac{19 \cos 37.46^\circ}{2.25 \times 2.3} = 2.9 \text{ KSI}$

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WHITING REQN. 79624 DATE 2-13-87  
 BY ASZ PAGE 4-19 OF 4  
 MJM 8-24-87

SECTION B-B (TENSION)

DIRECT  $\sigma_D = \frac{F_V}{A}$

BENDING  $\sigma_{BH} = \frac{M_H}{S} = \frac{F_H(2.15)}{S}$

$\sigma_{BV} = \frac{M_V}{S} = \frac{F_V \left( \frac{4.68}{2} - 0.375 + 2.563 \right)}{S}$

$\sigma_B = \sigma_D + \sigma_{BH} + \sigma_{BV}$

OBE  $\sigma_B = \frac{6.7}{2.25 \times 4.68} + \frac{1.9 \times 2.15}{\frac{2.25 \times 4.68^2}{6}} + \frac{6.7 \times 2.53}{\frac{2.25 \times 4.68^2}{6}} = 3.2 \text{ KSI}$

SSE  $\sigma_B = \frac{18.3}{2.25 \times 4.68} + \frac{5.2 \times 2.15}{\frac{2.25 \times 4.68^2}{6}} + \frac{18.3 \times 2.53}{\frac{2.25 \times 4.68^2}{6}} = 8.7 \text{ KSI}$

SECTION C-C (TENSION)

DIRECT  $\sigma_D = \frac{F_V}{A}$

BENDING  $\sigma_{BH} = \frac{F_H(0.23)}{S}$

$\sigma_{BV} = \frac{F_V \left( \frac{2.86}{2} + 0.563 \right)}{S}$

$\sigma_C = \sigma_D + \sigma_{BH} + \sigma_{BV}$

OBE  $\sigma_C = \frac{6.7}{2.25 \times 2.86} + \frac{1.9 \times 0.20}{\frac{2.25 \times 2.86^2}{6}} + \frac{6.7 \times 1.99}{\frac{2.25 \times 2.86^2}{6}} = 5.5 \text{ KSI}$

SSE  $\sigma_C = \frac{18.3}{2.25 \times 2.86} + \frac{5.2 \times 0.20}{\frac{2.25 \times 2.86^2}{6}} + \frac{18.3 \times 1.99}{\frac{2.25 \times 2.86^2}{6}} = 15.1 \text{ KSI}$

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WHITING REQN. 72604 DATE 8-3-57  
 BY ASZ PAGE 4-20 OF 3  
 MJM 8-24-97

SECTION A-A (TENSION)

$$\text{DIRECT } \sigma_D = \frac{FR \sin(53.42-30+14.04)}{A}$$

$$\text{BENDING } \sigma_{BH} = \frac{-F_H(0.69-0.20)}{S}$$

$$\sigma_{BV} = \frac{F_V(0.92+0.56)}{S}$$

$$\text{TENSILE } \sigma_A = \sigma_D + \sigma_{BH} + \sigma_{BV}$$

$$\text{OBE: } \sigma_A = \frac{6.9 \sin 37.46^\circ}{2.25 \times 2.3} - \frac{1.9 \times 0.49}{\frac{2.25 \times 2.3^2}{6}} + \frac{6.7 \times 1.48}{\frac{2.25 \times 2.3^2}{6}} = 5.3 \text{ ksi}$$

$$\text{SSA } \sigma_A = \frac{19 \sin 37.46^\circ}{2.25 \times 2.3} - \frac{5.2 \times 0.49}{\frac{2.25 \times 2.3^2}{6}} + \frac{18.3 \times 1.48}{\frac{2.25 \times 2.3^2}{6}} = 14.6 \text{ ksi}$$

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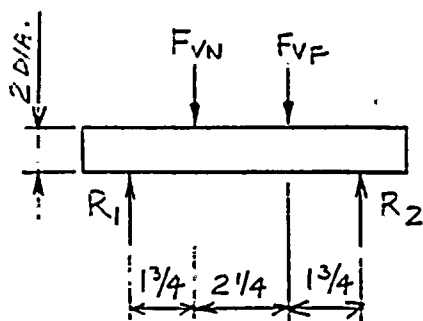
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WHITING REQ. 79604 DATE 2-13-2  
 BY ASZ PAGE 4-21 OF 2  
 MJM 8-24-87

LUG PIN



OBE  $P_{MAX} = 13.3 \text{ KIP}$

SSE  $P_{MAX} = 36.6 \text{ KIP}$

MATRL.: 1018 COLD FINISH

$\sigma_{YMIN} = 30 \text{ KSI}$

OBE  $\sigma_{ALL} = \frac{30}{1.5} = 20.0 \text{ KSI}$

$\tau_{ALL} = 0.6 \sigma_{ALL} = 12.0 \text{ KSI}$

SSE  $\sigma_{ALL} = \frac{30}{1.1} = 27.3 \text{ KSI}$

$\tau_{ALL} = 0.6 \sigma_{ALL} = 16.4 \text{ KSI}$

VERTICAL LOAD ON NEARSIDE PIN.

$$F_{VN} = \frac{(\frac{10}{2} + 1.875)F_V - (2 + 0.23 + 2.5)F_H}{10} = \frac{6.875(0.5 P_{MAX}) - 4.73(0.143 P_{MAX})}{10}$$

OBE  $F_{VN} = 3.7 \text{ KIP}$

SSE  $F_{VN} = 10.1 \text{ KIP}$

VERTICAL LOAD ON FAR SIDE PIN.

$$F_{VF} = \frac{(\frac{10}{2} - 1.875)F_V + 4.73F_H}{10} = \frac{3.125(0.5 P_{MAX}) + 4.73(0.143 P_{MAX})}{10}$$

OBE  $F_{VF} = 3.0 \text{ KIP}$

SSE  $F_{VF} = 8.2 \text{ KIP}$

WHITING REQ. 72604 DATE 2-2-2  
 BY ASZ PAGE 4-22 OF 12  
 MJM 8-24-87

$$R_1 = \frac{F_{VN}(5.75-1.75) + F_{VF}(1.75)}{5.75}$$

$$R_2 = \frac{F_{VN}(1.75) + F_{VF}(5.75-1.75)}{5.75}$$

OBE  $R_1 = 3.5 \text{ KIP}$   $R_2 = 3.2 \text{ KIP}$

SSE  $R_1 = 9.5 \text{ KIP}$   $R_2 = 8.8 \text{ KIP}$

$$\frac{\text{ARM}}{\text{DEPTH}} = \frac{1.75}{2} = 0.875 < 1.5$$

OK

∴ PROBABLE MODE OF FAILURE IS SHEAR

$$\text{SHEAR STRESS } \tau = \frac{4}{3} \frac{R_{MAX}}{A} = \frac{4}{3} \frac{R_1}{\frac{\pi 2^2}{4}}$$

OBE  $\tau = 1.5 \text{ KSI}$

SSE  $\tau = 4.0 \text{ KSI}$

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WHITING REQ. 79504 DATE 2-2-87  
 BY AJZ PAGE 4-23 OF 2  
 MJM 8-24-87

WELDS

ALLOWABLES

MTRL: ASTM-A36  $\sigma_{YMIN} = 36$  KSI

WELD MTRL: E70XX ELECTRODES FOR WELD TYPE A AND B,  $\sigma_{YMIN} = 57$  KSI

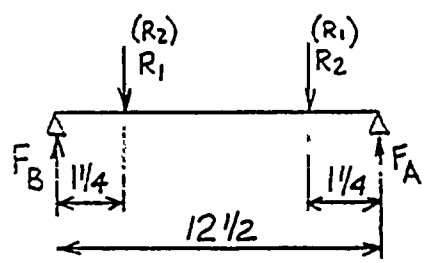
OBE  $\tau_{W,ALL} = \frac{57 \times 0.6}{1.5} = 22.8$  KSI  
 $= \frac{0.6 \times 36 \sqrt{2}}{1.5} = 20.4$  KSI  
 $\therefore \tau_{W,ALL} = 20.4$  KSI

SSE  $\tau_{W,ALL} = \frac{57 \times 0.6}{1.1} = 31.1$  KSI  
 $= \frac{0.6 \times 36 \sqrt{2}}{1.1} = 27.8$  KSI  
 $\therefore \tau_{W,ALL} = 27.8$  KSI

WELD MTRL: E60XX ELECTRODES FOR WELD TYPE C,  $\sigma_{YMIN} = 50$  KSI

OBE  $\tau_{W,ALL} = \frac{50 \times 0.6}{1.5} = 20$  KSI

SSE  $\tau_{W,ALL} = \frac{50 \times 0.6}{1.1} = 27.3$  KSI



MAX. FORCE ON WELD A, B

$F_{AMAX} = F_{BMAX} = \frac{R_1(11.25) + R_2(1.25)}{12.5}$

OBE  $F_{AMAX} = F_{BMAX} = 3.5$  KIP

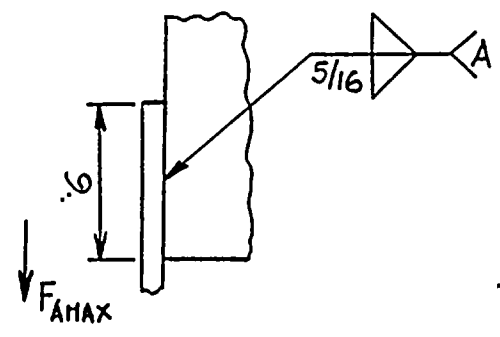
SSE  $F_{AMAX} = F_{BMAX} = 9.4$  KIP

WELD A

$\tau_A = \frac{F_{AMAX}}{A} = \frac{F_{AMAX}}{0.707 \times 0.3125(2 \times 6)}$

OBE  $\tau_A = 13$  KSI

SSE  $\tau_A = 3.5$  KSI





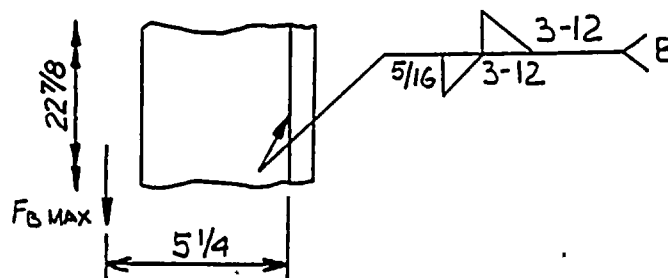
WHITING REQ. 79604 DATE 2. 4. 67  
 BY ASZ PAGE 4-24 OF 4  
 MJM 8-25-37

WELD B

$$\tau_B = \sqrt{\tau_Y^2 + \tau_Z^2} = \frac{F_{B \text{ MAX}}}{0.707(0.3125)\left(\frac{3}{12}\right)2} \sqrt{\left(\frac{5.25}{\frac{22.875^2}{6}}\right)^2 + \left(\frac{1}{22.875}\right)^2} = \frac{F_{B \text{ MAX}}}{1.48}$$

OBE  $\tau_B = 2.4 \text{ KSI}$

SSE  $\tau_B = 6.3 \text{ KSI}$

WELD C

MAX. FORCE ON WELD = P<sub>MAX</sub>

OBE 13.3 KIP

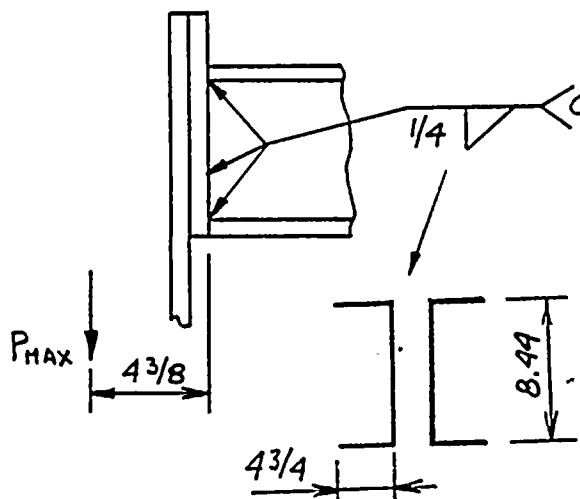
SSE 36.6 KIP (REF. PG. 4-16)

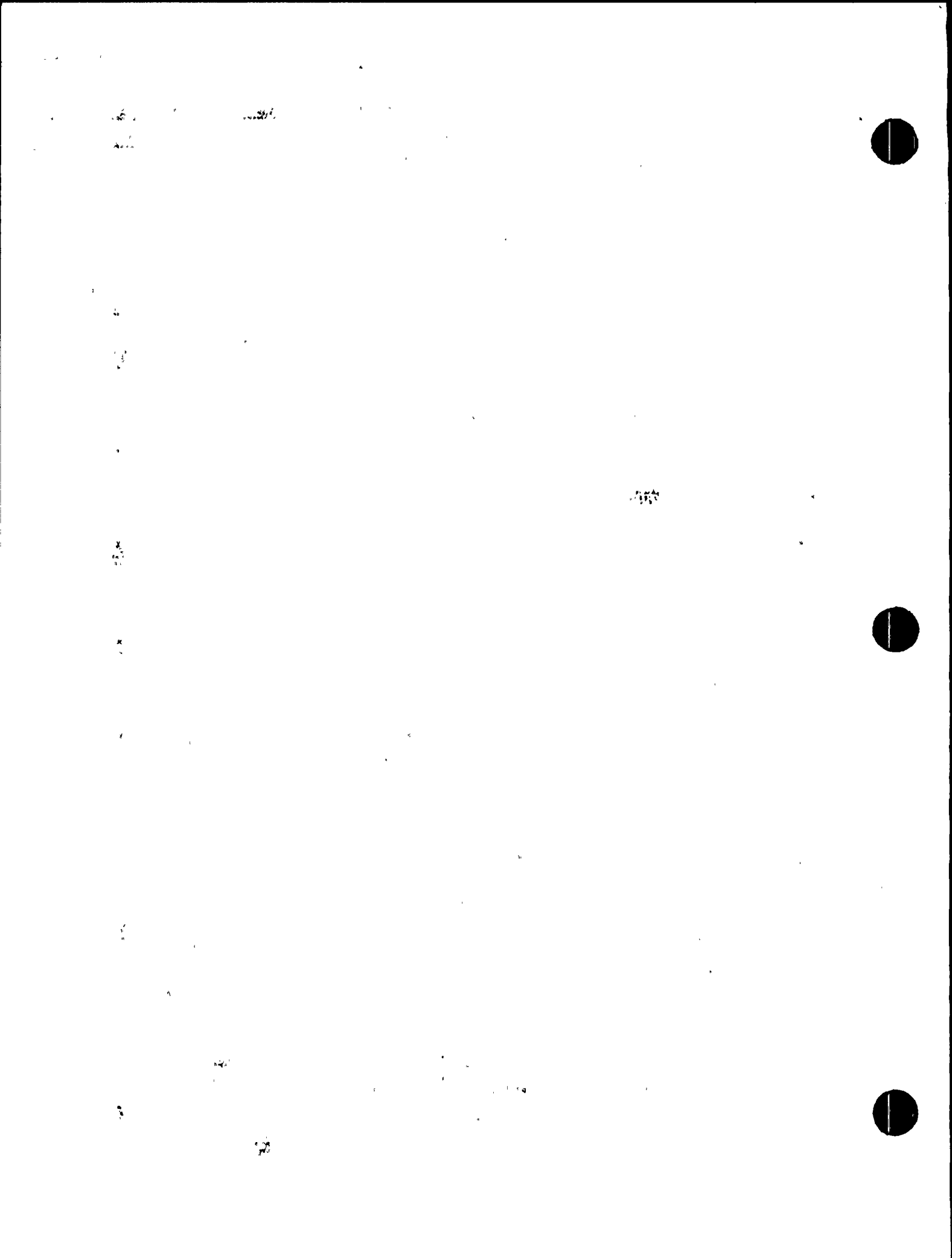
$$\tau_C = \sqrt{\tau_Y^2 + \tau_Z^2} = \frac{P_{\text{MAX}}}{0.707 \times 0.25} \sqrt{\left(\frac{4.38}{2(4.75 \times 8.44) + \frac{8.44^2}{3}}\right)^2 + \left(\frac{1}{2(2 \times 4.75 + 8.44)}\right)^2}$$

$$= \frac{P_{\text{MAX}}}{3.50}$$

OBE  $\tau_C = 3.8 \text{ KSI}$

SSE  $\tau_C = 10.5 \text{ KSI}$





WHITING REQ. 72604 DATE 2-2-57  
 BY ACZ PAGE 4-25 OF 42  
 MJM 8-24-87

BOLTS

$\frac{7}{8}$ " DIA. A325 4 REQ'D

$$G_{YMIN} = 92 \text{ KSI}$$

$$\text{OBE } G_{ALL} = \frac{92}{1.5} = 61.3 \text{ KSI}$$

$$\tau_{ALL} = 0.6 G_{ALL} = 36.8 \text{ KSI}$$

$$\text{SSE } G_{ALL} = \frac{92}{1.1} = 83.6 \text{ KSI}$$

$$\tau_{ALL} = 0.6 G_{ALL} = 50.2 \text{ KSI}$$

MAX. FORCE ON BOLTS =  $P_{MAX}$

OBE 13.3 KIP

SSE 36.6 KIP

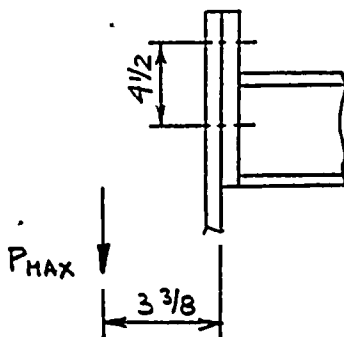
$$\text{SHEAR STRESS } \tau = \frac{P_{MAX}}{nA} = \frac{P_{MAX}}{4 \cdot \frac{\pi}{4} (0.875)^2}$$

$$\text{OBE } \tau = 5.5 \text{ KSI}$$

$$\text{SSE } \tau = 15.2 \text{ KSI}$$

$$\text{TENSILE STRESS } G = \frac{M}{S} = \frac{P_{MAX}(3.38)}{4A(2.25)} = \frac{P_{MAX}}{2.66}$$

$$S = \frac{I}{r} \approx \frac{4Ar^2}{r} = 4Ar$$



FOR  $\frac{7}{8}$  BOLTS (4 REQ'D)

$$\text{TENSILE AREA } A_T = 0.462 \text{ IN}^2$$

$$\text{SHANK AREA } A_S = 0.601 \text{ IN}^2$$



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WHITING REQ. 79604 DATE 3-14-87  
 BY ASZ PAGE 4-26 OF 112  
 MJM 8-24-87

IN THREADS

OBE  $\sigma = 10.8 \text{ KSI}$

SSE  $\sigma = 29.7 \text{ KSI}$

IN SHANK

OBE  $\sigma = 8.3 \text{ KSI}$

SSE  $\sigma = 22.8 \text{ KSI}$

COMBINED IN SHANK

$$\tau_{\text{COMB}} = \sqrt{\left(\frac{\sigma}{2}\right)^2 + \tau^2}$$

$$\text{OBE } \tau_{\text{COMB}} = \sqrt{\left(\frac{8.3}{2}\right)^2 + 5.5^2} = 6.9 \text{ KSI}$$

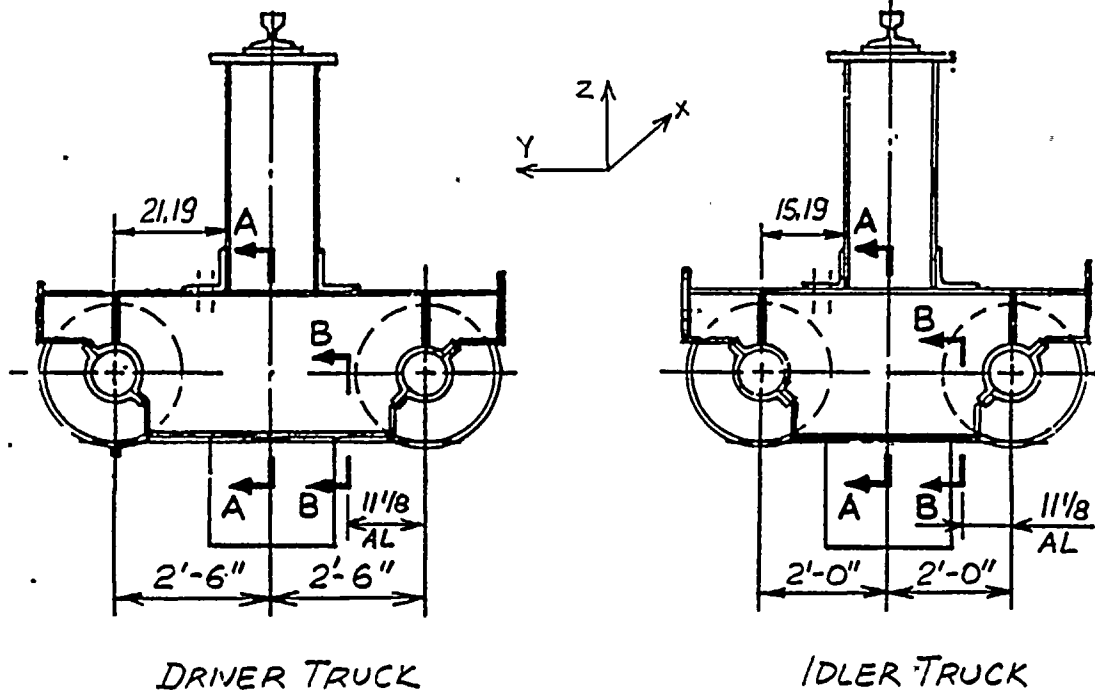
$$\text{SSE } \tau_{\text{COMB}} = \sqrt{\left(\frac{22.8}{2}\right)^2 + 15.2^2} = 19.0 \text{ KSI}$$

Vertical text or markings on the left side of the page, possibly bleed-through from the reverse side.



WHITING REQ. 72504 DATE 3-13-37  
 BY ASZ PAGE 2-27 OF 12  
 M.J.M. 3-26-37

BRIDGE TRUCKS



DRIVER TRUCK

IDLER TRUCK

MTRL: ASTM-A36

$G_{YMIN} = 36 \text{ KSI}$

OBE  $G_{ALL} = \frac{G_{YMIN}}{1.5} = 24 \text{ KSI}$

$T_{ALL} = 0.6 G_{ALL} = 14.4 \text{ KSI}$

SSE  $G_{ALL} = \frac{G_{YMIN}}{1.1} = 32.7 \text{ KSI}$

$T_{ALL} = 0.6 G_{ALL} = 19.6 \text{ KSI}$

MAX. LOADINGS PER TABLES 4-4 AND 4-5.

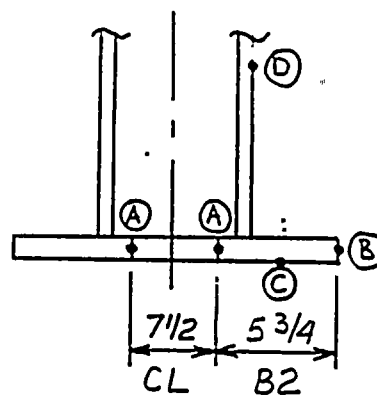
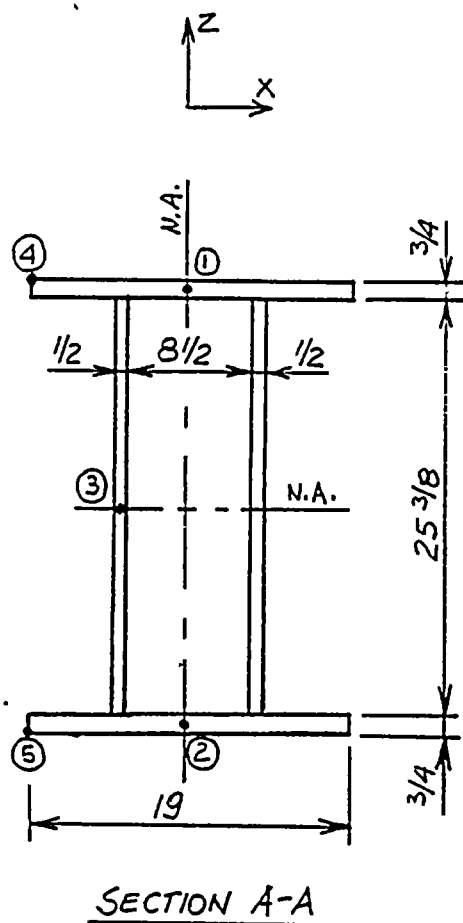
OBE  $F_x = 5.88 \text{ KIP}$      $F_y = 44.9 \text{ KIP}$      $F_z = 179.6 \text{ KIP}$

$F_x = 5.88 \text{ KIP}$      $F_y = 0 \text{ KIP}$      $F_z = 138.1 \text{ KIP}$

SSE  $F_x = 11.1 \text{ KIP}$      $F_y = 64.2 \text{ KIP}$      $F_z = 256.8 \text{ KIP}$

$F_x = 11.1 \text{ KIP}$      $F_y = 0 \text{ KIP}$      $F_z = 195.2 \text{ KIP}$

WHITING REQ. 79604 DATE 5-2-57  
 BY ASZ PAGE 4-28 OF 12  
 MJM 8-26-57



SECTION A-A WAS SIMPLIFIED FOR COMPUTER PROGRAM PURPOSE.

ACTUAL SECTION CONSISTS OF TWO W27x94 BEAMS WITH FLANGES CUT OFF FROM ONE SIDE 1/2 IN. AND WELDED TOGETHER (TOTAL FLANGE LENGTH EQUALS 19 IN.)

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AEP 50T BRIDGE DRIVER TRUCK OBE

WHITING REQN. 79604 DATE 8-25-37  
 BY RGG PAGE 4-29 OF 112  
 MJM 8-26-37

TRUCK SIZE , TOP PLT = 0.75 X 19.00 IN.  
 BOT PLT = 0.75 X 19.00 IN.  
 WEB PLT = 0.50 X 25.38 IN.  
 DISTANCE BETWEEN WEB PLTS = 8.50 IN.

TRUCK WHEEL BASE = 42.38 IN.  
 COEF. OF FRICTION = 0.250

RAIL TO CENTER OF TWIST = 15.938 IN.

**\*\* FORCES IN THE GLOBAL COORDINATE SYSTEM \*\***

FX = 5880.0 LBS.  
 FY = 44900.0 LBS.  
 FZ = 179600.0 LBS.

**\*\* SECTION PROPERTIES \*\***

AREA IN SQ. IN. ----- 53.9  
 VERTICAL MOMENT OF INERTIA----- 6225.8  
 VERTICAL SECTION MODULUS----- 463.3  
 VERTICAL CENTER OF GRAVITY----- 13.4  
 HORIZ. MOMENT OF INERTIA----- 1371.7  
 HORIZ. SECTION MODULUS----- 144.4

**\*\* MAXIMUM DEFLECTIONS IN INCHES \*\***

VERTICAL BENDING DEFL. DUE TO FZ----- 0.00305  
 HORIZ. BENDING DEFL. DUE TO FX----- 0.00045  
 VERTICAL SHEAR DEFL. DUE TO FZ----- 0.01419  
 HORIZ. SHEAR DEFL. DUE TO FX----- 0.00065

**\*\* TRUCK STRESSES IN PSI \*\* AT POINT (N) SECTION A-A**

BENDING STRESS TOP FLANGE DUE TO FX----- 862.9 (4)  
 BENDING STRESS BOT FLANGE DUE TO FX----- 862.9 (5)  
 BENDING STRESS TOP FLANGE DUE TO FY----- 1544.5 (1)(4)  
 BENDING STRESS BOT FLANGE DUE TO FY----- 1544.5 (2)(5)  
 BENDING STRESS TOP FLANGE DUE TO FZ----- 8214.1 (1)(4)  
 BENDING STRESS BOT FLANGE DUE TO FZ----- 8214.1 (2)(5)  
 BENDING STRESS ON WEB DUE TO FX----- 431.4 (3)  
 BENDING STRESS ON WEB DUE TO FY----- 0.0 (3)  
 TRANSVERSE SHEAR ON FLANGE DUE TO FX----- 356.6 (1)(2)  
 TRANSVERSE SHEAR ON WEB DUE TO FZ----- 7701.9 (3)  
 TORSIONAL SHEAR ON FLANGE DUE TO FX----- 265.7 (1)(2)  
 TORSIONAL SHEAR ON FLANGE DUE TO FZ----- 0.0 (3)  
 DIRECT TENSILE STRESS DUE TO FY----- 833.4 (ALL)  
 TORSIONAL SHEAR ON WEB DUE TO FX----- 398.6 (3)  
 TORSIONAL SHEAR ON WEB DUE TO FZ----- 0.0 (3)

**\*\* ALLOWABLE BENDING = 24000.0 \*\* ALLOWABLE SHEAR = 14400.0**

BEND. STRESS IN TOP FLG. DUE TO FY & FZ-- 9758.6 (1)  
 BEND. STRESS IN BOT. FLG. DUE TO FY & FZ-- 9758.6 (2)  
 BENDING STRESS IN WEB DUE TO FX & FY----- 431.4 (3)  
 TOTAL TORSIONAL SHEAR STRESS IN FLANGES-- 265.7 (1)(2)  
 TOTAL TORSIONAL SHEAR STRESS IN WEB----- 398.6 (3)  
 MAX TENSILE STRESS IN TOP FLANGE----- (PSI) 11454.9 (4)  
 MAX TENSILE STRESS IN BOT FLANGE----- (PSI) 11454.9 (5)  
 MAX SHEAR STRESS-CENTER TOP FLANGE----- 5332.4 (1)  
 MAX SHEAR STRESS-CENTER BOT FLANGE----- 5332.4 (2)  
 MAX SHEAR STRESS-CENTER WEB----- (PSI) 8124.7 (3)

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WHITING REQ. 79654 DATE 8-25-87  
 BY RGG PAGE 4-30 OF 112  
 MJM 8-26-87

## AEP 50T BRIDGE DRIVER TRUCK OBE

## \*\* ANALYSIS OF TRUCK END DUE TO TORSION \*\*

ADDED DIMENSIONS - B2 = 5.750 IN.  
 AL = 11.125 IN.  
 CL = 7.500 IN.

TORSIONAL RESISTANCE----- 6.4 IN\*\*4  
 SHEAR CENTER FROM CENTER OF TOP PLT-- 10.217 IN.  
 WARPING CONSTANT----- 51145.2 IN\*\*6.  
 WARPING MOMENT----- 93712.5 IN-LBS.  
 ANGLE OF TWIST - 2ND DERIVATIVE----- 0.3396E-06  
 3RD DERIVATIVE----- 0.6108E-07

## \*\* TRUCK END STRESSES IN PSI \*\*

TENSILE STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. A----- 712.0

TENSILE STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. B----- 2544.0

SHEAR STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. C  
 AT A DISTANCE (Y) 1.943 IN. FROM  
 THE INSIDE FACE OF THE WEB----- 307.9

SHEAR STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. D  
 AT A DISTANCE (Z) 10.717 IN. FROM  
 THE TOP FLANGE----- 811.2

TRANSVERSE SHEAR STRESS IN FLG DUE TO FX- 412.6  
 TRANSVERSE SHEAR STRESS IN WEB DUE TO FZ- 6682.8

MAX SHEAR STRESS IN WEB @ PT. D----- 7494.0  
 MAX SHEAR STRESS IN BOT FLANGE----- 1308.7

\*\* STRESS CALC. CALLED 2 TIMES

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AEP 50T BRIDGE IDLER TRUCK OBE

WHITING REON. 79604 DATE 8-25-87  
 BY RGG PAGE 4-31 OF 42  
 MJM 8-26-87

TRUCK SIZE , TOP PLT = 0.75 X 19.00 IN.  
 BOT PLT = 0.75 X 19.00 IN.  
 WEB PLT = 0.50 X 25.38 IN.  
 DISTANCE BETWEEN WEB PLTS = 8.50 IN.

TRUCK WHEEL BASE = 30.38 IN.  
 COEF. OF FRICTION = 0.250

RAIL TO CENTER OF TWIST = 15.938 IN.

**\*\* FORCES IN THE GLOBAL COORDINATE SYSTEM \*\***

FX = 5880.0 LBS.  
 FY = 0.0 LBS.  
 FZ = 138100.0 LBS.

**\*\* SECTION PROPERTIES \*\***

AREA IN SQ. IN. ----- 53.9  
 VERTICAL MOMENT OF INERTIA----- 6225.8  
 VERTICAL SECTION MODULUS----- 463.3  
 VERTICAL CENTER OF GRAVITY----- 13.4  
 HORIZ. MOMENT OF INERTIA----- 1371.7  
 HORIZ. SECTION MODULUS----- 144.4

**\*\* MAXIMUM DEFLECTIONS IN INCHES \*\***

VERTICAL BENDING DEFL. DUE TO FZ----- 0.00086  
 HORIZ. BENDING DEFL. DUE TO FX----- 0.00017  
 VERTICAL SHEAR DEFL. DUE TO FZ----- 0.00782  
 HORIZ. SHEAR DEFL. DUE TO FX----- 0.00047

**\*\* TRUCK STRESSES IN PSI \*\* AT POINT (N) SECTION A-A**

BENDING STRESS TOP FLANGE DUE TO FX----- 618.6 (4)  
 BENDING STRESS BOT FLANGE DUE TO FX----- 618.6 (5)  
 BENDING STRESS TOP FLANGE DUE TO FY----- 0.0 (1)(4)  
 BENDING STRESS BOT FLANGE DUE TO FY----- 0.0 (2)(5)  
 BENDING STRESS TOP FLANGE DUE TO FZ----- 4527.7 (1)(4)  
 BENDING STRESS BOT FLANGE DUE TO FZ----- 4527.7 (2)(5)  
 BENDING STRESS ON WEB DUE TO FX----- 309.3 (3)  
 BENDING STRESS ON WEB DUE TO FY----- 0.0 (9)  
 TRANSVERSE SHEAR ON FLANGE DUE TO FX----- 356.6 (1)(2)  
 TRANSVERSE SHEAR ON WEB DUE TO FZ----- 5921.9 (3)  
 TORSIONAL SHEAR ON FLANGE DUE TO FX----- 265.7 (1)(2)  
 TORSIONAL SHEAR ON FLANGE DUE TO FZ----- 0.0 (3)  
 DIRECT TENSILE STRESS DUE TO FY----- 0.0 (ALL)  
 TORSIONAL SHEAR ON WEB DUE TO FX----- 398.6 (3)  
 TORSIONAL SHEAR ON WEB DUE TO FZ----- 0.0 (3)

**\*\* ALLOWABLE BENDING = 24000.0 \*\* ALLOWABLE SHEAR = 14400.0**

BEND. STRESS IN TOP FLG. DUE TO FY & FZ-- 4527.7 (1)  
 BEND. STRESS IN BOT. FLG. DUE TO FY & FZ-- 4527.7 (2)  
 BENDING STRESS IN WEB DUE TO FX & FY----- 309.3 (3)  
 TOTAL TORSIONAL SHEAR STRESS IN FLANGES-- 265.7 (1)(2)  
 TOTAL TORSIONAL SHEAR STRESS IN WEB----- 398.6 (3)  
 MAX TENSILE STRESS IN TOP FLANGE----- (psi) 5146.2 (4)  
 MAX TENSILE STRESS IN BOT FLANGE----- (psi) 5146.2 (5)  
 MAX SHEAR STRESS-CENTER TOP FLANGE----- 2347.8 (1)  
 MAX SHEAR STRESS-CENTER BOT FLANGE----- 2347.8 (2)  
 MAX SHEAR STRESS-CENTER WEB----- (psi) 6322.4 (3)

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WHITING REQN. 79604 DATE 8-25-71  
 BY RGG PAGE 4-32 OF 12  
 MJM 8-26-87

## AEP 50T BRIDGE IDLER TRUCK OBE

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**\*\* ANALYSIS OF TRUCK END DUE TO TORSION \*\***


---

ADDED DIMENSIONS - B2 = 5.750 IN.  
 AL = 11.125 IN.  
 CL = 7.500 IN.

TORSIONAL RESISTANCE----- 6.4 IN\*\*4  
 SHEAR CENTER FROM CENTER OF TOP PLT-- 10.217 IN.  
 WARPING CONSTANT----- 51145.9 IN\*\*4  
 WARPING MOMENT----- 93712.5 IN-LBS.  
 ANGLE OF TWIST - 2ND DERIVATIVE----- 0.3396E-06  
 3RD DERIVATIVE----- 0.6108E-07

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**\*\* TRUCK END STRESSES IN PSI \*\***


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TENSILE STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. A----- 712.0

TENSILE STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. B----- 2544.0

SHEAR STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. C;  
 AT A DISTANCE (Y) 1.943 IN. FROM  
 THE INSIDE FACE OF THE WEB----- 307.9

SHEAR STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. D;  
 AT A DISTANCE (Z) 10.717 IN. FROM  
 THE TOP FLANGE----- 811.2

TRANSVERSE SHEAR STRESS IN FLG DUE TO FX- 412.6  
 TRANSVERSE SHEAR STRESS IN WEB DUE TO FZ- 5138.6

MAX SHEAR STRESS IN WEB @ PT. D----- 5949.8  
 MAX SHEAR STRESS IN BOT FLANGE----- 1308.7

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**\*\* STRESS CALC. CALLED 2 TIMES**


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WHITING REON. 79604 DATE 8-25-87  
 BY PGG PAGE 4-33 OF 112  
 MJM 8-26-87

AEP 50T BRIDGE DRIVER TRUCK SSE

TRUCK SIZE , TOP PLT = 0.75 X 19.00 IN.  
 BOT PLT = 0.75 X 19.00 IN.  
 WEB PLT = 0.50 X 25.38 IN.  
 DISTANCE BETWEEN WEB PLTS = 8.50 IN.

TRUCK WHEEL BASE = 42.38 IN.  
 COEF. OF FRICTION = 0.250

RAIL TO CENTER OF TWIST = 15.938 IN.

\*\* FORCES IN THE GLOBAL COORDINATE SYSTEM \*\*

FX = 11100.0 LBS.  
 FY = 64200.0 LBS.  
 FZ = 256800.0 LBS.

\*\* SECTION PROPERTIES \*\*

AREA IN SQ. IN. ----- 53.9  
 VERTICAL MOMENT OF INERTIA----- 6225.8  
 VERTICAL SECTION MODULUS----- 463.3  
 VERTICAL CENTER OF GRAVITY----- 13.4  
 HORIZ. MOMENT OF INERTIA----- 1371.7  
 HORIZ. SECTION MODULUS----- 144.4

\*\* MAXIMUM DEFLECTIONS IN INCHES \*\*

VERTICAL BENDING DEFL. DUE TO FZ----- 0.00436  
 HORIZ. BENDING DEFL. DUE TO FX----- 0.00086  
 VERTICAL SHEAR DEFL. DUE TO FZ----- 0.02029  
 HORIZ. SHEAR DEFL. DUE TO FX----- 0.00124

\*\* TRUCK STRESSES IN PSI \*\* AT POINT (N) SECTION A-A

BENDING STRESS TOP FLANGE DUE TO FX----- 1628.9 (4)  
 BENDING STRESS BOT FLANGE DUE TO FX----- 1628.9 (5)  
 BENDING STRESS TOP FLANGE DUE TO FY----- 2208.4 (1)(4)  
 BENDING STRESS BOT FLANGE DUE TO FY----- 2208.4 (2)(5)  
 BENDING STRESS TOP FLANGE DUE TO FZ----- 11744.9 (7)(4)  
 BENDING STRESS BOT FLANGE DUE TO FZ----- 11744.9 (2)(5)  
 BENDING STRESS ON WEB DUE TO FX----- 814.5 (3)  
 BENDING STRESS ON WEB DUE TO FY----- 0.0 (3)  
 TRANSVERSE SHEAR ON FLANGE DUE TO FX----- 673.1 (1)(2)  
 TRANSVERSE SHEAR ON WEB DUE TO FZ----- 11012.0 (3)  
 TORSIONAL SHEAR ON FLANGE DUE TO FX----- 501.6 (1)(2)  
 TORSIONAL SHEAR ON FLANGE DUE TO FZ----- 0.0 (3)  
 DIRECT TENSILE STRESS DUE TO FY----- 1191.6 (ALL)  
 TORSIONAL SHEAR ON WEB DUE TO FX----- 752.4 (3)  
 TORSIONAL SHEAR ON WEB DUE TO FZ----- 0.0 (3)

\*\* ALLOWABLE BENDING = 32700.0 \*\* ALLOWABLE SHEAR = 19600.0

BEND. STRESS IN TOP FLG. DUE TO FY & FZ-- 13953.3 (1)  
 BEND. STRESS IN BOT. FLG. DUE TO FY & FZ-- 13953.3 (2)  
 BENDING STRESS IN WEB DUE TO FX & FY----- 814.5 (3)  
 TOTAL TORSIONAL SHEAR STRESS IN FLANGES-- 501.6 (1)(2)  
 TOTAL TORSIONAL SHEAR STRESS IN WEB----- 752.4 (3)  
 MAX TENSILE STRESS IN TOP FLANGE----- (psi) 16773.8 (4)  
 MAX TENSILE STRESS IN BOT FLANGE----- (psi) 16773.8 (5)  
 MAX SHEAR STRESS-CENTER TOP FLANGE----- 7663.0 (1)  
 MAX SHEAR STRESS-CENTER BOT FLANGE----- 7663.0 (2)  
 MAX SHEAR STRESS-CENTER WEB----- (psi) 11807.0 (3)

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WHITING REON. 79604 DATE 8-25-87  
 BY REG PAGE 4-34 OF 112  
 MJM 8-26-87

AEP 50T BRIDGE DRIVER TRUCK SSE

**\*\* ANALYSIS OF TRUCK END DUE TO TORSION \*\***

ADDED DIMENSIONS - B2 = 5.750 IN.  
 AL = 11.125 IN.  
 CL = 7.500 IN.

TORSIONAL RESISTANCE----- 6.4 IN\*\*4  
 SHEAR CENTER FROM CENTER OF TOP PLT-- 10.217 IN.  
 WARPING CONSTANT----- 51145.9 IN\*\*6  
 WARPING MOMENT----- 176906.3 IN-LBS.  
 ANGLE OF TWIST - 2ND DERIVATIVE----- 0.6410E-06  
 3RD DERIVATIVE----- 0.1153E-06

**\*\* TRUCK END STRESSES IN PSI \*\***

TENSILE STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. A----- 1344.1  
 TENSILE STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. B----- 4802.5  
 SHEAR STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. C:  
 AT A DISTANCE (Y) 1.943 IN. FROM  
 THE INSIDE FACE OF THE WEB----- 581.2  
 SHEAR STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. D:  
 AT A DISTANCE (Z) 10.717 IN. FROM  
 THE TOP FLANGE----- 1531.3  
 TRANSVERSE SHEAR STRESS IN FLG DUE TO FX- 778.9  
 TRANSVERSE SHEAR STRESS IN WEB DUE TO FZ- 9555.3  
 MAX SHEAR STRESS IN WEB @ PT. D----- 11086.7  
 MAX SHEAR STRESS IN BOT FLANGE----- 2470.6

**\*\* STRESS CALC. CALLED 2 TIMES**

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AEP 50T BRIDGE-IDLER TRUCK SSE

WHITING REQN. 79604 DATE 8-25-87  
BY RGG PAGE 4-35 OF 112  
MJM 8-26-87

TRUCK SIZE, TOP PLT = 0.75 X 19.00 IN.  
 BOT PLT = 0.75 X 19.00 IN.  
 WEB PLT = 0.50 X 25.38 IN.  
 DISTANCE BETWEEN WEB PLTS = 8.50 IN.

TRUCK WHEEL BASE = 30.38 IN.  
 COEF. OF FRICTION = 0.250

RAIL TO CENTER OF TWIST = 15.938 IN.

## \*\* FORCES IN THE GLOBAL COORDINATE SYSTEM \*\*

FX = 11100.0 LBS.  
 FY = 0.0 LBS.  
 FZ = 195200.0 LBS.

## \*\* SECTION PROPERTIES \*\*

AREA IN SQ. IN. ----- 53.9  
 VERTICAL MOMENT OF INERTIA----- 6225.8  
 VERTICAL SECTION MODULUS----- 463.3  
 VERTICAL CENTER OF GRAVITY----- 13.4  
 HORIZ. MOMENT OF INERTIA----- 1371.7  
 HORIZ. SECTION MODULUS----- 144.4

## \*\* MAXIMUM DEFLECTIONS IN INCHES \*\*

VERTICAL BENDING DEFL. DUE TO FZ----- 0.00122  
 HORIZ. BENDING DEFL. DUE TO FX----- 0.00032  
 VERTICAL SHEAR DEFL. DUE TO FZ----- 0.01106  
 HORIZ. SHEAR DEFL. DUE TO FX----- 0.00089

## \*\* TRUCK STRESSES IN PSI \*\* AT POINT (N) SECTION A-A

BENDING STRESS TOP FLANGE DUE TO FX----- 1167.7 (4)  
 BENDING STRESS BOT FLANGE DUE TO FX----- 1167.7 (5)  
 BENDING STRESS TOP FLANGE DUE TO FY----- 0.0 (1)(4)  
 BENDING STRESS BOT FLANGE DUE TO FY----- 0.0 (2)(5)  
 BENDING STRESS TOP FLANGE DUE TO FZ----- 6399.7 (1)(4)  
 BENDING STRESS BOT FLANGE DUE TO FZ----- 6399.7 (2)(5)  
 BENDING STRESS ON WEB DUE TO FX----- 583.8 (3)  
 BENDING STRESS ON WEB DUE TO FY----- 0.0 (3)  
 TRANSVERSE SHEAR ON FLANGE DUE TO FX----- 673.1 (1)(2)  
 TRANSVERSE SHEAR ON WEB DUE TO FZ----- 8370.5 (3)  
 TORSIONAL SHEAR ON FLANGE DUE TO FX----- 501.6 (1)(2)  
 TORSIONAL SHEAR ON FLANGE DUE TO FZ----- 0.0 (3)  
 DIRECT TENSILE STRESS DUE TO FY----- 0.0 (ALL)  
 TORSIONAL SHEAR ON WEB DUE TO FX----- 752.4 (3)  
 TORSIONAL SHEAR ON WEB DUE TO FZ----- 0.0 (3)

\*\* ALLOWABLE BENDING = 32700.0 \*\* ALLOWABLE SHEAR = 19600.0 \*\*

BEND. STRESS IN TOP FLG. DUE TO FY & FZ-- 6399.7 (1)  
 BEND. STRESS IN BOT. FLG. DUE TO FY & FZ-- 6399.7 (2)  
 BENDING STRESS IN WEB DUE TO FX & FY----- 583.8 (3)  
 TOTAL TORSIONAL SHEAR STRESS IN FLANGES-- 501.6 (1)(2)  
 TOTAL TORSIONAL SHEAR STRESS IN WEB----- 752.4 (3)  
 MAX TENSILE STRESS IN TOP FLANGE----- (PSI) 7567.4 (4)  
 MAX TENSILE STRESS IN BOT FLANGE----- (PSI) 7567.4 (5)  
 MAX SHEAR STRESS-CENTER TOP FLANGE----- 3408.7 (1)  
 MAX SHEAR STRESS-CENTER BOT FLANGE----- 3408.7 (2)  
 MAX SHEAR STRESS-CENTER WEB----- (PSI) 9127.5 (3)

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WHITING REQ. 79604 DATE 8-25-87  
 BY RGG PAGE 4-36 OF 12  
 MJM 8-26-87

## AEP 50T BRIDGE IDLER TRUCK SSE

-----  
 \*\* ANALYSIS OF TRUCK END DUE TO TORSION \*\*  
 -----

ADDED DIMENSIONS - B2 = 5.750 IN.  
 AL = 11.125 IN.  
 CL = 7.500 IN.

TORSIONAL RESISTANCE----- 6.4 IN\*\*4  
 SHEAR CENTER FROM CENTER OF TOP PLT-- 10.217 IN.  
 WARPING CONSTANT----- 51145.9 IN\*\*6  
 WARPING MOMENT----- 176906.3 IN-LBS.  
 ANGLE OF TWIST - 2ND DERIVATIVE----- 0.6410E-06  
 3RD DERIVATIVE----- 0.1153E-06

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 \*\* TRUCK END STRESSES IN PSI \*\*  
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TENSILE STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. A----- 1344.1

TENSILE STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. B----- 4802.5

SHEAR STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. C  
 AT A DISTANCE (Y) 1.943 IN. FROM  
 THE INSIDE FACE OF THE WEB----- 581.2

SHEAR STRESS DUE TO WARPING RIGIDITY  
 THROUGHOUT THE THICKNESS @ PT. D  
 AT A DISTANCE (Z) 10.717 IN. FROM  
 THE TOP FLANGE----- 1531.3

TRANSVERSE SHEAR STRESS IN FLG DUE TO FX- 778.9  
 TRANSVERSE SHEAR STRESS IN WEB DUE TO FZ- 7263.3

MAX SHEAR STRESS IN WEB @ PT. D----- 8794.6  
 MAX SHEAR STRESS IN BOT FLANGE----- 2470.6

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 \*\* STRESS CALC. CALLED 2 TIMES  
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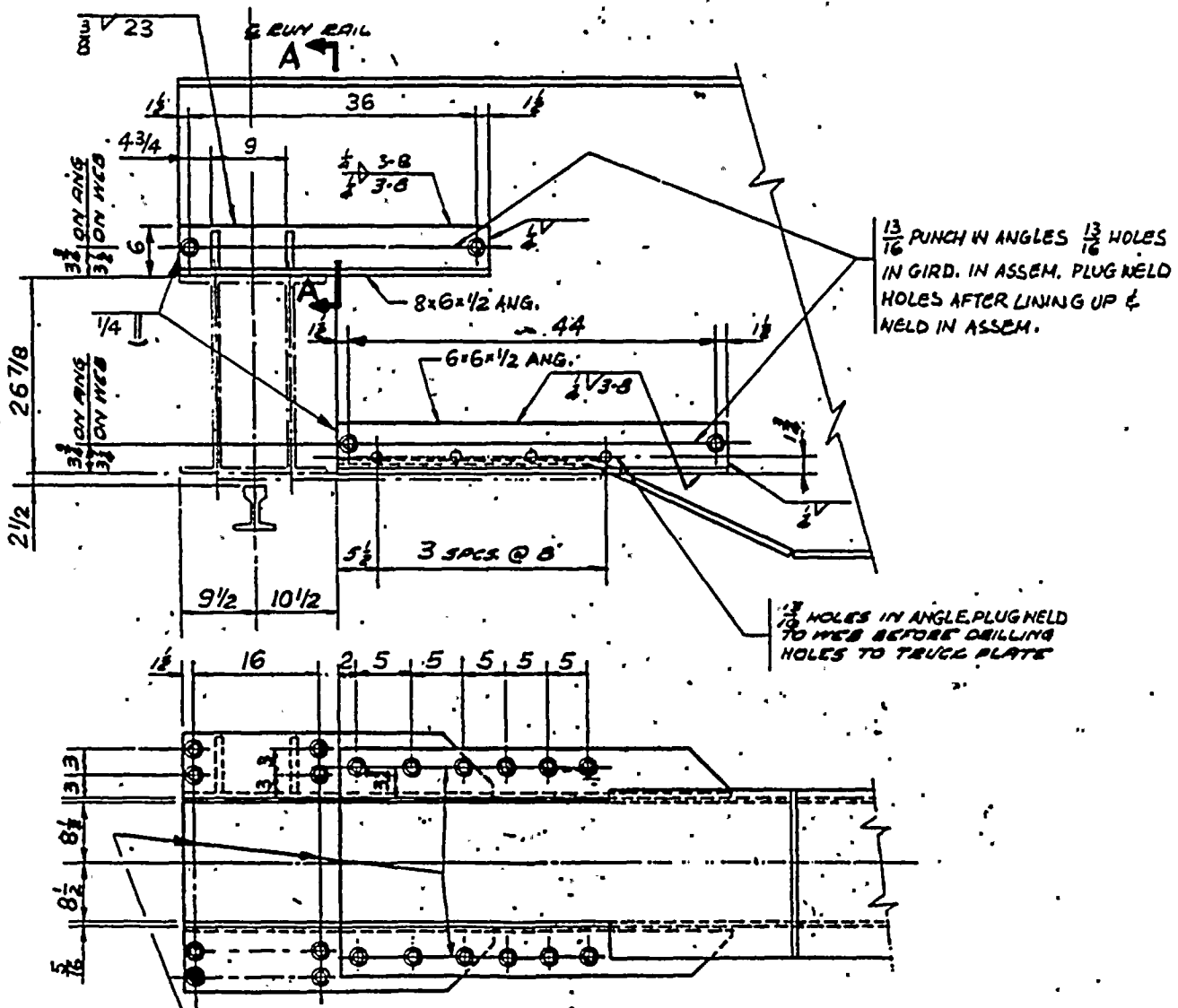
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WHITING REQN. 79604 DATE 6-22-87  
BY ASZ PAGE 4-37 OF 112  
MJM 8-25-87

# GIRDER TO TRUCK CONNECTION.



7/8 A325 BOLTS

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WHITING REQ. 79604 DATE 6-22-87  
 BY ASZ PAGE 4-38 OF 112  
 MJM 8-25-87

ALLOWABLES7/8" BOLTS

MTRL: A325

$$S_{YMIN} = 92 \text{ KSI}$$

$$\text{OBE } \sigma_{ALL} = \frac{S_{YMIN}}{1.5} = 61.3 \text{ KSI}$$

$$\text{SSE } \sigma_{ALL} = \frac{S_{YMIN}}{1.1} = 83.6 \text{ KSI}$$

$$\tau_{ALL} = 0.6 \sigma_{ALL} = 36.8 \text{ KSI}$$

$$\tau_{ALL} = 0.6 \sigma_{ALL} = 50.2 \text{ KSI}$$

WELDS (FILLET THRU THROAT)BASE MTRL: ASTM-A36  $S_{YMIN} = 36 \text{ KSI}$ WELD MTRL: MIN. E60XX ELECTRODES  $S_{YMIN} = 50 \text{ KSI}$ 

$$\text{OBE } \tau_{W.ALL} = \frac{S_{YMIN.WELD}}{1.5} \cdot 0.6 = 20.0 \text{ KSI}$$

$$= \frac{0.6 \sqrt{2} S_{YMIN.BASE}}{1.5} = 20.4 \text{ KSI}$$

$$\therefore \tau_{W.ALL} = 20.0 \text{ KSI}$$

$$\text{SSE } \tau_{W.ALL} = \frac{0.6 S_{YMIN.WELD}}{1.1} = 27.3 \text{ KSI}$$

$$= \frac{0.6 \sqrt{2} S_{YMIN.BASE}}{1.1} = 27.8 \text{ KSI}$$

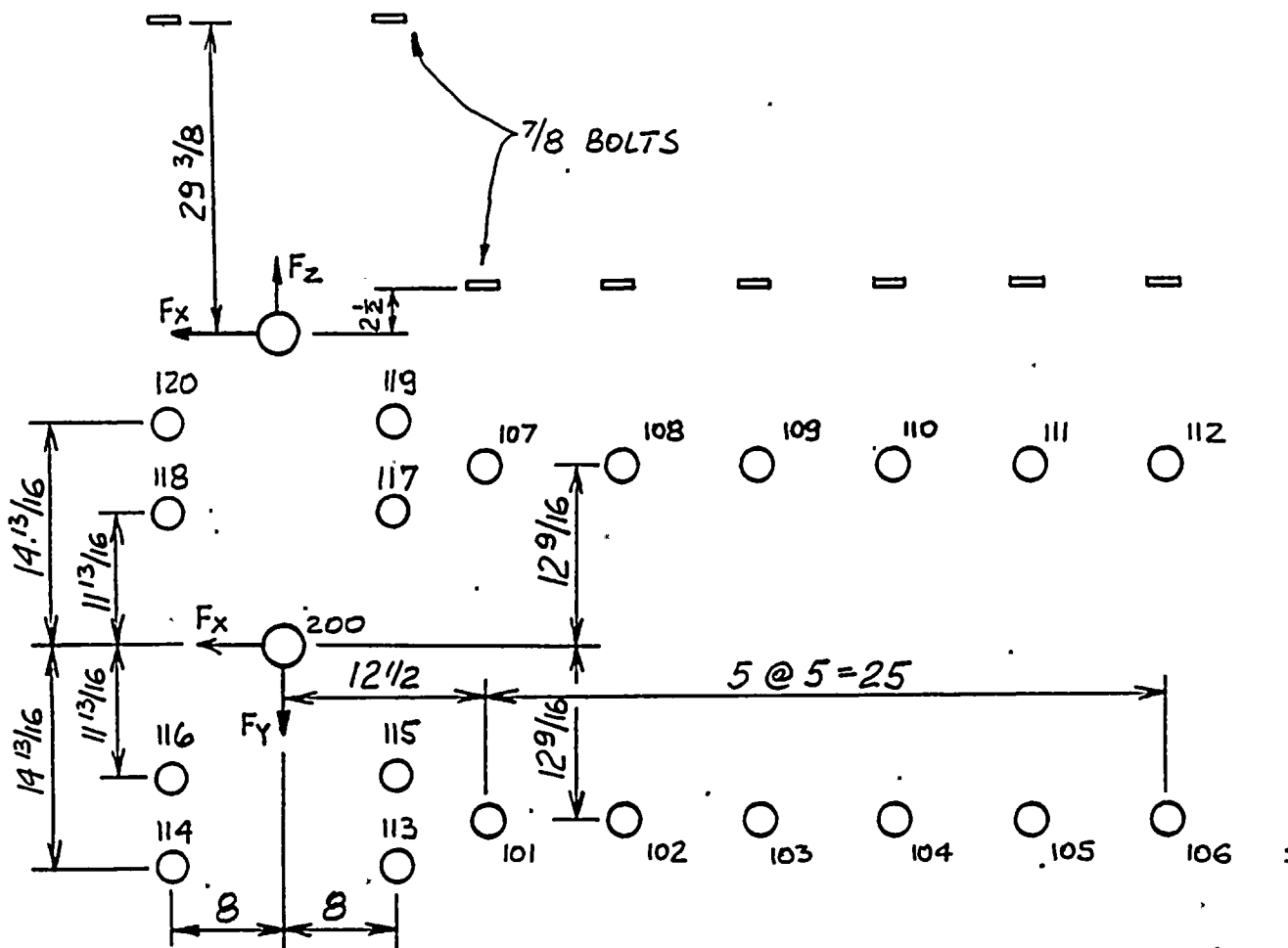
$$\therefore \tau_{W.ALL} = 27.3 \text{ KSI}$$

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WHITING REQN. 79604 DATE 5-22-87  
 BY ASZ PAGE 4-39 OF 112  
 MJM 8-25-87

BOLTED CONNECTION



$F_z$  LOAD IS TRANSFERRED IN BEARING OF WEBS ON TRUCK. OTHER LOADS ARE TRANSFERRED IN SHEAR OF BOLTS. BOLTS ARE NOT IN TENSION.

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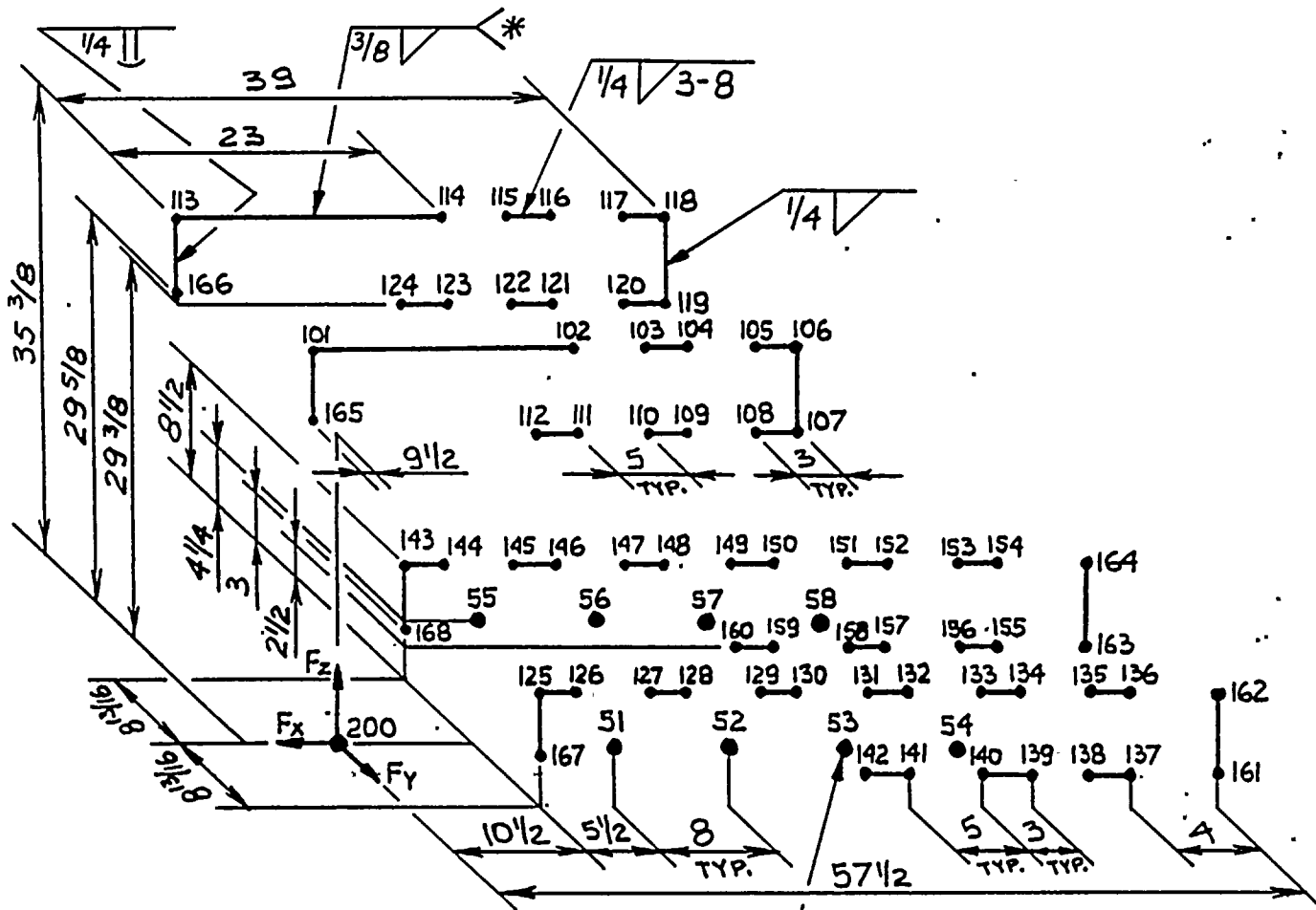
WHITING REQN. 79604 DATE 6-23-87  
 BY ASZ PAGE 4-40 OF 112  
 MJM 8-25-87

MAX. LOADINGS PER TABLE 4-2 AND 4-3

OBE  $F_x = 11.8$  KIP  $F_y = 49.8$  KIP  $F_z = 250.2$  KIP  $M_x = 3267$  IN. KIP

SSE  $F_x = 22.2$  KIP  $F_y = 78.8$  KIP  $F_z = 339.4$  KIP  $M_x = 5384$  IN. KIP

WELDED CONNECTION.



\* 5/16 EFFECTIVE BECAUSE OF WEB SIZE.

13/16 DIA. □

1/4 SQUARE WELDS WERE TREATED AS 1/4 FILLET WELDS.

PLUG WELDS WERE OMITTED, WHICH IS CONSERVATIVE.

BECAUSE WELDS ARE LOADED BY BOLTS. THEY ARE CONSIDERED INEFFECTIVE IN Z DIRECTION (BOLT TENSION).

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* OBE

\*\*\*\*\* WORST CASE ANALYSIS SUMMARY \*\*\*\*\*

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* OBE

LOAD STEP	DIRECTION		
	X	Y	Z
1			
TRANSLATED FORCES	11.80	49.80	0.00
TRANSLATED MOMENTS	-2607.15	-156.35	747.00
NUMBER OF FORCE DEFINITION NODES	1		
MAXIMUM ABSOLUTE STRESS ON ELEMENT 14 = 28.1 KSI			
COMPARISON FACTOR MATCH ON ELEMENT 14 = 1.3081508			

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* OBE

LOAD STEP	DIRECTION		
	X	Y	Z
2			
TRANSLATED FORCES	11.80	-49.80	0.00
TRANSLATED MOMENTS	2607.15	-156.35	-747.00
NUMBER OF FORCE DEFINITION NODES	1		
MAXIMUM ABSOLUTE STRESS ON ELEMENT 20 = 28.1 KSI			
COMPARISON FACTOR MATCH ON ELEMENT 20 = 1.3081508			

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* OBE

LOAD STEP	DIRECTION		
	X	Y	Z
3			
TRANSLATED FORCES	-11.80	49.80	0.00
TRANSLATED MOMENTS	-2607.15	156.35	747.00
NUMBER OF FORCE DEFINITION NODES	1		
MAXIMUM ABSOLUTE STRESS ON ELEMENT 20 = 28.1 KSI			
COMPARISON FACTOR MATCH ON ELEMENT 20 = 1.3081508			

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* OBE

LOAD STEP	DIRECTION		
	X	Y	Z
4			
TRANSLATED FORCES	-11.80	-49.80	0.00
TRANSLATED MOMENTS	2607.15	156.35	-747.00
NUMBER OF FORCE DEFINITION NODES	1		
MAXIMUM ABSOLUTE STRESS ON ELEMENT 14 = 28.1 KSI			
COMPARISON FACTOR MATCH ON ELEMENT 14 = 1.3081508			

\*\*\*\*\* WORST CASE ANALYSIS COMPLETE \*\*\*\*\*

WORST CASE OCCURED DURING LOAD STEP 1

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* OBE

\*\*\*\*\* SYSTEM PROPERTIES \*\*\*\*\*

LOAD STEP..... 1 DIRECTION, LOCATION OR AXIS

	X	Y	Z
CENTROIDS, X AXIS.....		0.00	13.25
Y AXIS.....	0.00		13.25
Z AXIS.....	-15.00	-0.00	
SHEAR AREAS.....	12.03	12.03	0.00
POLAR MOMENTS OF INERTIA..	2084.70	2084.70	4640.13
TRANSLATED FORCES.....	11.80	49.80	0.00
TRANSLATED MOMENTS.....	-2607.15	-156.35	747.00

NUMBER OF FORCE DEFINITION NODES... 1

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

DIRECT STRESSES ARE EQUALLY DISTRIBUTED THROUGHOUT SHEAR AREA  
 DRX 1.0 DRY 4.1 DRZ 0.0

2-D POINT ELEMENT 1	NODE 101	AREA 0.601	DIAMETER 0.875
STRESS AT NODE	8.9	ALLOWABLE 36.8	
STRESS EXPANSION FOR NODE 101			
0.8	-2.0	-13.4	0.4 0.0 0.0
FORCES AT NODE 101	FX	-0. FY	-5. FZ 0.
2-D POINT ELEMENT 2	NODE 102	AREA 0.601	DIAMETER 0.875
STRESS AT NODE	9.7	ALLOWABLE 36.8	
STRESS EXPANSION FOR NODE 102			
0.8	-2.0	-13.4	-0.4 0.0 0.0
FORCES AT NODE 102	FX	-0. FY	-6. FZ 0.
2-D POINT ELEMENT 3	NODE 103	AREA 0.601	DIAMETER 0.875
STRESS AT NODE	10.5	ALLOWABLE 36.8	
STRESS EXPANSION FOR NODE 103			
0.8	-2.0	-13.4	-1.2 0.0 0.0
FORCES AT NODE 103	FX	-0. FY	-6. FZ 0.
2-D POINT ELEMENT 4	NODE 104	AREA 0.601	DIAMETER 0.875
STRESS AT NODE	11.3	ALLOWABLE 36.8	
STRESS EXPANSION FOR NODE 104			
0.8	-2.0	-13.4	-2.0 0.0 0.0
FORCES AT NODE 104	FX	-0. FY	-7. FZ 0.
2-D POINT ELEMENT 5	NODE 105	AREA 0.601	DIAMETER 0.875
STRESS AT NODE	12.1	ALLOWABLE 36.8	
STRESS EXPANSION FOR NODE 105			
0.8	-2.0	-13.4	-2.8 0.0 0.0
FORCES AT NODE 105	FX	-0. FY	-7. FZ 0.

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* OBE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

61	2-D POINT ELEMENT	6	NODE 106	AREA	0.601	DIAMETER	0.875
62	STRESS AT NODE	12.9	ALLOWABLE	36.8			
63	STRESS EXPANSION FOR NODE 106						
64	0.8	-2.0	-13.4	-3.6	0.0	0.0	
65	FORCES AT NODE 106	FX	-0.	FY	-8.	FZ	0.
66	2-D POINT ELEMENT	7	NODE 107	AREA	0.601	DIAMETER	0.875
67	STRESS AT NODE	9.7	ALLOWABLE	36.8			
68	STRESS EXPANSION FOR NODE 107						
69	0.8	2.0	-13.4	0.4	0.0	0.0	
70	FORCES AT NODE 107	FX	2.	FY	-5.	FZ	0.
71	2-D POINT ELEMENT	8	NODE 108	AREA	0.601	DIAMETER	0.875
72	STRESS AT NODE	10.4	ALLOWABLE	36.8			
73	STRESS EXPANSION FOR NODE 108						
74	0.8	2.0	-13.4	-0.4	0.0	0.0	
75	FORCES AT NODE 108	FX	2.	FY	-6.	FZ	0.
76	2-D POINT ELEMENT	9	NODE 109	AREA	0.601	DIAMETER	0.875
77	STRESS AT NODE	11.2	ALLOWABLE	36.8			
78	STRESS EXPANSION FOR NODE 109						
79	0.8	2.0	-13.4	-1.2	0.0	0.0	
80	FORCES AT NODE 109	FX	2.	FY	-6.	FZ	0.
81	2-D POINT ELEMENT	10	NODE 110	AREA	0.601	DIAMETER	0.875
82	STRESS AT NODE	11.9	ALLOWABLE	36.8			
83	STRESS EXPANSION FOR NODE 110						
84	0.8	2.0	-13.4	-2.0	0.0	0.0	
85	FORCES AT NODE 110	FX	2.	FY	-7.	FZ	0.
86	2-D POINT ELEMENT	11	NODE 111	AREA	0.601	DIAMETER	0.875
87	STRESS AT NODE	12.7	ALLOWABLE	36.8			
88	STRESS EXPANSION FOR NODE 111						
89	0.8	2.0	-13.4	-2.8	0.0	0.0	
90	FORCES AT NODE 111	FX	2.	FY	-7.	FZ	0.
91	2-D POINT ELEMENT	12	NODE 112	AREA	0.601	DIAMETER	0.875
92	STRESS AT NODE	13.5	ALLOWABLE	36.8			
93	STRESS EXPANSION FOR NODE 112						
94	0.8	2.0	-13.4	-3.6	0.0	0.0	
95	FORCES AT NODE 112	FX	2.	FY	-8.	FZ	0.
96	2-D POINT ELEMENT	13	NODE 113	AREA	0.601	DIAMETER	0.875
97	STRESS AT NODE	25.6	ALLOWABLE	36.8			
98	STRESS EXPANSION FOR NODE 113						
99	-1.2	-2.4	20.2	1.1	0.0	0.0	
100	FORCES AT NODE 113	FX	-2.	FY	15.	FZ	0.
101	2-D POINT ELEMENT	14	NODE 114	AREA	0.601	DIAMETER	0.875
102	STRESS AT NODE	28.1	ALLOWABLE	36.8			
103	STRESS EXPANSION FOR NODE 114						
104	-1.2	-2.4	20.2	3.7	0.0	0.0	
105	FORCES AT NODE 114	FX	-2.	FY	17.	FZ	0.

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* OBE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

2-D POINT ELEMENT 15 NODE 115 AREA 0.601 DIAMETER 0.875

STRESS AT NODE 25.5 ALLOWABLE 36.8

STRESS EXPANSION FOR NODE 115

-1.2 -1.9 20.2 1.1 0.0 0.0

FORCES AT NODE 115 FX -1. FY 15. FZ 0.

2-D POINT ELEMENT 16 NODE 116 AREA 0.601 DIAMETER 0.875

STRESS AT NODE 28.1 ALLOWABLE 36.8

STRESS EXPANSION FOR NODE 116

-1.2 -1.9 20.2 3.7 0.0 0.0

FORCES AT NODE 116 FX -1. FY 17. FZ 0.

2-D POINT ELEMENT 17 NODE 117 AREA 0.601 DIAMETER 0.875

STRESS AT NODE 25.5 ALLOWABLE 36.8

STRESS EXPANSION FOR NODE 117

-1.2 1.9 20.2 1.1 0.0 0.0

FORCES AT NODE 117 FX 1. FY 15. FZ 0.

2-D POINT ELEMENT 18 NODE 118 AREA 0.601 DIAMETER 0.875

STRESS AT NODE 28.1 ALLOWABLE 36.8

STRESS EXPANSION FOR NODE 118

-1.2 1.9 20.2 3.7 0.0 0.0

FORCES AT NODE 118 FX 1. FY 17. FZ 0.

2-D POINT ELEMENT 19 NODE 119 AREA 0.601 DIAMETER 0.875

STRESS AT NODE 25.5 ALLOWABLE 36.8

STRESS EXPANSION FOR NODE 119

-1.2 2.4 20.2 1.1 0.0 0.0

FORCES AT NODE 119 FX 1. FY 15. FZ 0.

2-D POINT ELEMENT 20 NODE 120 AREA 0.601 DIAMETER 0.875

STRESS AT NODE 28.1 ALLOWABLE 36.8

STRESS EXPANSION FOR NODE 120

-1.2 2.4 20.2 3.7 0.0 0.0

FORCES AT NODE 120 FX 1. FY 17. FZ 0.

FORCE DEFINITION NODE DIRECT ELEMENT 21 NODE 200

FX 11.80 ; FY 49.80 ; FZ 0.00

MX -3267.00 ; MY 0.00 ; MZ 0.00

MAXIMUM ABSOLUTE STRESS ON ELEMENT 14 = 28.1 KSI

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* OBE

\*\*\*\*\* REPORT SUMMARY, FOR A 4 LOAD STEP ANALYSIS \*\*\*\*\*

WORST CASE ANALYSIS.....NO  
 ELEMENT STRESSES PRINTED....YES, STRESS SUMMATION MODE WAS DIRECT

ELEMENT STATISTICS

TYPE	DESCRIPTION	NUMBER USED
1	2-DIMENSIONAL POINT ELEMENT.....	20
2	FORCE DEFINITION NODE - DIRECT.....	1

TOTAL NUMBER OF ELEMENTS = 21, NODES = 21

SYSTEM PROPERTIES

LOAD STEP.....	1	DIRECTION, LOCATION OR AXIS		
		X	Y	Z
CENTROIDS, X AXIS.....			0.00	13.25
Y AXIS.....		0.00		13.25
Z AXIS.....		-15.00	-0.00	
SHEAR AREAS.....		12.03	12.03	0.00
POLAR MOMENTS OF INERTIA..		2084.70	2084.70	4640.13
TRANSLATED FORCES.....		11.80	49.80	0.00
TRANSLATED MOMENTS.....		-2607.15	-156.35	747.00

MAXIMUM ELEMENT STRESSES

DESCRIPTION	NUMBER		ELEMENT	NODE	STRESS	ALLOWABLE	COMPARISON FACTOR
	USED						
2-D POINT ELEMENT	20		14	114	28.1 KSI	36.8 KSI	1.3082

\*\*\*\*\* JOB COMPLETED \*\*\*\*\*

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* SSE

\*\*\*\*\* WORST CASE ANALYSIS SUMMARY \*\*\*\*\*

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* SSE

LOAD STEP	DIRECTION		
	X	Y	Z
TRANSLATED FORCES	22.20	78.80	0.00
TRANSLATED MOMENTS	-4339.90	-294.15	1182.00
NUMBER OF FORCE DEFINITION NODES	1		
MAXIMUM ABSOLUTE STRESS ON ELEMENT 14	=	46.2 KSI	
COMPARISON FACTOR MATCH ON ELEMENT 14	=	1.0872481	

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* SSE

LOAD STEP	DIRECTION		
	X	Y	Z
TRANSLATED FORCES	22.20	-78.80	0.00
TRANSLATED MOMENTS	4339.90	-294.15	-1182.00
NUMBER OF FORCE DEFINITION NODES	1		
MAXIMUM ABSOLUTE STRESS ON ELEMENT 20	=	46.2 KSI	
COMPARISON FACTOR MATCH ON ELEMENT 20	=	1.0872481	

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* SSE

LOAD STEP	DIRECTION		
	X	Y	Z
TRANSLATED FORCES	-22.20	78.80	0.00
TRANSLATED MOMENTS	-4339.90	294.15	1182.00
NUMBER OF FORCE DEFINITION NODES	1		
MAXIMUM ABSOLUTE STRESS ON ELEMENT 20	=	46.2 KSI	
COMPARISON FACTOR MATCH ON ELEMENT 20	=	1.0872481	

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* SSE

LOAD STEP	DIRECTION		
	X	Y	Z
TRANSLATED FORCES	-22.20	-78.80	0.00
TRANSLATED MOMENTS	4339.90	294.15	-1182.00
NUMBER OF FORCE DEFINITION NODES	1		
MAXIMUM ABSOLUTE STRESS ON ELEMENT 14	=	46.2 KSI	
COMPARISON FACTOR MATCH ON ELEMENT 14	=	1.0872481	

\*\*\*\*\* WORST CASE ANALYSIS COMPLETE \*\*\*\*\*

WORST CASE OCCURED DURING LOAD STEP 1

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* SSE

\*\*\*\*\* SYSTEM PROPERTIES \*\*\*\*\*

LOAD STEP.....	1	DIRECTION, LOCATION OR AXIS		
		X	Y	Z
CENTROIDS, X AXIS.....			0.00	13.25
Y AXIS.....		0.00		13.25
Z AXIS.....		-15.00	-0.00	
SHEAR AREAS.....		12.03	12.03	0.00
POLAR MOMENTS OF INERTIA..		2084.70	2084.70	4640.13
TRANSLATED FORCES.....		22.20	78.80	0.00
TRANSLATED MOMENTS.....		-4339.90	-294.15	1182.00
NUMBER OF FORCE DEFINITION NODES...	1			

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

DIRECT STRESSES ARE EQUALLY DISTRIBUTED THROUGHOUT SHEAR AREA									
DRX	1.8	DRY	6.6	DRZ	0.0				
2-D POINT ELEMENT	1	NODE 101	AREA	0.601	DIAMETER	0.875			
STRESS AT NODE	15.2	ALLOWABLE	50.2						
STRESS EXPANSION FOR NODE 101	1.5	-3.2	-22.4	0.6	0.0	0.0			
FORCES AT NODE 101	FX	0.	FY	-9.	FZ	0.			
2-D POINT ELEMENT	2	NODE 102	AREA	0.601	DIAMETER	0.875			
STRESS AT NODE	16.5	ALLOWABLE	50.2						
STRESS EXPANSION FOR NODE 102	1.5	-3.2	-22.4	-0.6	0.0	0.0			
FORCES AT NODE 102	FX	0.	FY	-10.	FZ	0.			
2-D POINT ELEMENT	3	NODE 103	AREA	0.601	DIAMETER	0.875			
STRESS AT NODE	17.7	ALLOWABLE	50.2						
STRESS EXPANSION FOR NODE 103	1.5	-3.2	-22.4	-1.9	0.0	0.0			
FORCES AT NODE 103	FX	0.	FY	-11.	FZ	0.			
2-D POINT ELEMENT	4	NODE 104	AREA	0.601	DIAMETER	0.875			
STRESS AT NODE	19.0	ALLOWABLE	50.2						
STRESS EXPANSION FOR NODE 104	1.5	-3.2	-22.4	-3.2	0.0	0.0			
FORCES AT NODE 104	FX	0.	FY	-11.	FZ	0.			
2-D POINT ELEMENT	5	NODE 105	AREA	0.601	DIAMETER	0.875			
STRESS AT NODE	20.3	ALLOWABLE	50.2						
STRESS EXPANSION FOR NODE 105	1.5	-3.2	-22.4	-4.5	0.0	0.0			
FORCES AT NODE 105	FX	0.	FY	-12.	FZ	0.			





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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* SSE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

2-D POINT ELEMENT	6	NODE 106	AREA	0.601	DIAMETER	0.875
STRESS AT NODE	21.6	ALLOWABLE	50.2			
STRESS EXPANSION FOR NODE 106						
	1.5	-3.2	-22.4	-5.7	0.0	0.0
FORCES AT NODE 106	FX	0.	FY	-13.	FZ	0.
2-D POINT ELEMENT	7	NODE 107	AREA	0.601	DIAMETER	0.875
STRESS AT NODE	16.5	ALLOWABLE	50.2			
STRESS EXPANSION FOR NODE 107						
	1.5	3.2	-22.4	0.6	0.0	0.0
FORCES AT NODE 107	FX	4.	FY	-9.	FZ	0.
2-D POINT ELEMENT	8	NODE 108	AREA	0.601	DIAMETER	0.875
STRESS AT NODE	17.7	ALLOWABLE	50.2			
STRESS EXPANSION FOR NODE 108						
	1.5	3.2	-22.4	-0.6	0.0	0.0
FORCES AT NODE 108	FX	4.	FY	-10.	FZ	0.
2-D POINT ELEMENT	9	NODE 109	AREA	0.601	DIAMETER	0.875
STRESS AT NODE	18.9	ALLOWABLE	50.2			
STRESS EXPANSION FOR NODE 109						
	1.5	3.2	-22.4	-1.9	0.0	0.0
FORCES AT NODE 109	FX	4.	FY	-11.	FZ	0.
2-D POINT ELEMENT	10	NODE 110	AREA	0.601	DIAMETER	0.875
STRESS AT NODE	20.1	ALLOWABLE	50.2			
STRESS EXPANSION FOR NODE 110						
	1.5	3.2	-22.4	-3.2	0.0	0.0
FORCES AT NODE 110	FX	4.	FY	-11.	FZ	0.
2-D POINT ELEMENT	11	NODE 111	AREA	0.601	DIAMETER	0.875
STRESS AT NODE	21.3	ALLOWABLE	50.2			
STRESS EXPANSION FOR NODE 111						
	1.5	3.2	-22.4	-4.5	0.0	0.0
FORCES AT NODE 111	FX	4.	FY	-12.	FZ	0.
2-D POINT ELEMENT	12	NODE 112	AREA	0.601	DIAMETER	0.875
STRESS AT NODE	22.5	ALLOWABLE	50.2			
STRESS EXPANSION FOR NODE 112						
	1.5	3.2	-22.4	-5.7	0.0	0.0
FORCES AT NODE 112	FX	4.	FY	-13.	FZ	0.
2-D POINT ELEMENT	13	NODE 113	AREA	0.601	DIAMETER	0.875
STRESS AT NODE	42.1	ALLOWABLE	50.2			
STRESS EXPANSION FOR NODE 113						
	-2.3	-3.8	33.6	1.8	0.0	0.0
FORCES AT NODE 113	FX	-3.	FY	25.	FZ	0.
2-D POINT ELEMENT	14	NODE 114	AREA	0.601	DIAMETER	0.875
STRESS AT NODE	46.2	ALLOWABLE	50.2			
STRESS EXPANSION FOR NODE 114						
	-2.3	-3.8	33.6	5.9	0.0	0.0
FORCES AT NODE 114	FX	-3.	FY	28.	FZ	0.

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* SSE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

2-D POINT ELEMENT 15 NODE 115 AREA 0.601 DIAMETER 0.875

STRESS AT NODE 42.0 ALLOWABLE 50.2

STRESS EXPANSION FOR NODE 115

-2.3 -3.0 33.6 1.8 0.0 0.0

FORCES AT NODE 115 FX -2. FY 25. FZ 0.

2-D POINT ELEMENT 16 NODE 116 AREA 0.601 DIAMETER 0.875

STRESS AT NODE 46.1 ALLOWABLE 50.2

STRESS EXPANSION FOR NODE 116

-2.3 -3.0 33.6 5.9 0.0 0.0

FORCES AT NODE 116 FX -2. FY 28. FZ 0.

2-D POINT ELEMENT 17 NODE 117 AREA 0.601 DIAMETER 0.875

STRESS AT NODE 42.0 ALLOWABLE 50.2

STRESS EXPANSION FOR NODE 117

-2.3 3.0 33.6 1.8 0.0 0.0

FORCES AT NODE 117 FX 2. FY 25. FZ 0.

2-D POINT ELEMENT 18 NODE 118 AREA 0.601 DIAMETER 0.875

STRESS AT NODE 46.1 ALLOWABLE 50.2

STRESS EXPANSION FOR NODE 118

-2.3 3.0 33.6 5.9 0.0 0.0

FORCES AT NODE 118 FX 2. FY 28. FZ 0.

2-D POINT ELEMENT 19 NODE 119 AREA 0.601 DIAMETER 0.875

STRESS AT NODE 42.0 ALLOWABLE 50.2

STRESS EXPANSION FOR NODE 119

-2.3 3.8 33.6 1.8 0.0 0.0

FORCES AT NODE 119 FX 2. FY 25. FZ 0.

2-D POINT ELEMENT 20 NODE 120 AREA 0.601 DIAMETER 0.875

STRESS AT NODE 46.1 ALLOWABLE 50.2

STRESS EXPANSION FOR NODE 120

-2.3 3.8 33.6 5.9 0.0 0.0

FORCES AT NODE 120 FX 2. FY 28. FZ 0.

FORCE DEFINITION NODE DIRECT ELEMENT 21 NODE 200

FX 22.20 ; FY 78.80 ; FZ 0.00

MX -5384.00 ; MY 0.00 ; MZ 0.00

MAXIMUM ABSOLUTE STRESS ON ELEMENT 14 = 46.2 KSI

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* BOLTS \* SSE

\*\*\*\*\* REPORT SUMMARY, FOR A 4 LOAD STEP ANALYSIS \*\*\*\*\*

WORST CASE ANALYSIS.....NO  
 ELEMENT STRESSES PRINTED....YES, STRESS SUMMATION MODE WAS DIRECT

ELEMENT STATISTICS

TYPE	DESCRIPTION	NUMBER USED
1	2-DIMENSIONAL POINT ELEMENT.....	20
2	FORCE DEFINITION NODE - DIRECT.....	1

TOTAL NUMBER OF ELEMENTS = 21, NODES = 21

SYSTEM PROPERTIES

LOAD STEP..... 1 DIRECTION, LOCATION OR AXIS

	X	Y	Z
CENTROIDS, X AXIS.....		0.00	13.25
Y AXIS.....	0.00		13.25
Z AXIS.....	-15.00	-0.00	
SHEAR AREAS.....	12.03	12.03	0.00
POLAR MOMENTS OF INERTIA..	2084.70	2084.70	4640.13
TRANSLATED FORCES.....	22.20	78.80	0.00
TRANSLATED MOMENTS.....	-4339.90	-294.15	1182.00

MAXIMUM ELEMENT STRESSES

DESCRIPTION	NUMBER USED	ELEMENT	NODE	STRESS	ALLOWABLE	COMPARISON FACTOR
2-D POINT ELEMENT	20	14	114	46.2 ksi	50.2 ksi	1.0872

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* OBE

\*\*\*\*\* WORST CASE ANALYSIS SUMMARY \*\*\*\*\*

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* OBE

LOAD STEP.....	1	DIRECTION		
		X	Y	Z
TRANSLATED FORCES.....		11.80	49.80	0.00
TRANSLATED MOMENTS.....		-2146.55	-265.49	995.28
NUMBER OF FORCE DEFINITION NODES...		1		

MAXIMUM ABSOLUTE STRESS ON ELEMENT 38 = 7.9 KSI  
COMPARISON FACTOR MATCH ON ELEMENT 38 = 2.5354438

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* OBE

LOAD STEP.....	2	DIRECTION		
		X	Y	Z
TRANSLATED FORCES.....		11.80	-49.80	0.00
TRANSLATED MOMENTS.....		2146.55	-265.49	-995.28
NUMBER OF FORCE DEFINITION NODES...		1		

MAXIMUM ABSOLUTE STRESS ON ELEMENT 37 = 7.9 KSI  
COMPARISON FACTOR MATCH ON ELEMENT 37 = 2.5354452

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* OBE

LOAD STEP.....	3	DIRECTION		
		X	Y	Z
TRANSLATED FORCES.....		-11.80	49.80	0.00
TRANSLATED MOMENTS.....		-2146.55	265.49	995.28
NUMBER OF FORCE DEFINITION NODES...		1		

MAXIMUM ABSOLUTE STRESS ON ELEMENT 37 = 7.9 KSI  
COMPARISON FACTOR MATCH ON ELEMENT 37 = 2.5354433

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* OBE

LOAD STEP.....	4	DIRECTION		
		X	Y	Z
TRANSLATED FORCES.....		-11.80	-49.80	0.00
TRANSLATED MOMENTS.....		2146.55	265.49	-995.28
NUMBER OF FORCE DEFINITION NODES...		1		

MAXIMUM ABSOLUTE STRESS ON ELEMENT 38 = 7.9 KSI  
COMPARISON FACTOR MATCH ON ELEMENT 38 = 2.5354452

\*\*\*\*\* WORST CASE ANALYSIS COMPLETE \*\*\*\*\*

WORST CASE OCCURED DURING LOAD STEP 3

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LAEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* DBE

\*\*\*\*\* SYSTEM PROPERTIES \*\*\*\*\*

LOAD STEP	3	DIRECTION, LOCATION OR AXIS		
		X	Y	Z
CENTROIDS, X AXIS.....			0.00	22.50
Y AXIS.....		0.00		22.50
Z AXIS.....		-17.97	-0.00	
SHEAR AREAS.....		33.23	33.23	0.00
POLAR MOMENTS OF INERTIA..		6297.12	6297.12	15729.29
TRANSLATED FORCES.....		-11.80	49.80	0.00
TRANSLATED MOMENTS.....		-2146.55	265.49	995.28
NUMBER OF FORCE DEFINITION NODES...	1			

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

DIRECT STRESSES ARE EQUALLY DISTRIBUTED THROUGHOUT SHEAR AREA							
DRX	-0.4	DRY	1.5	DRZ	0.0		
3-D LINE ELEMENT	1	SIZE	0.313	LENGTH	23.000	AREA	5.082
STRESS AT NODES 101,102			7.8,		6.3	ALLOWABLE	20.0
STRESS EXPANSION FOR NODE 101							
0.5	-0.6		4.4		1.9	0.0	0.0
STRESS EXPANSION FOR NODE 102							
0.5	-0.6		4.4		0.4	0.0	0.0
3-D LINE ELEMENT	2	SIZE	0.313	LENGTH	23.000	AREA	5.082
STRESS AT NODES 113,114			7.8,		6.3	ALLOWABLE	20.0
STRESS EXPANSION FOR NODE 113							
0.5	0.6		4.4		1.9	0.0	0.0
STRESS EXPANSION FOR NODE 114							
0.5	0.6		4.4		0.4	0.0	0.0
3-D LINE ELEMENT	3	SIZE	0.250	LENGTH	3.000	AREA	0.530
STRESS AT NODES 103,104			6.0,		5.8	ALLOWABLE	20.0
STRESS EXPANSION FOR NODE 103							
0.5	-0.6		4.4		0.1	0.0	0.0
STRESS EXPANSION FOR NODE 104							
0.5	-0.6		4.4		-0.1	0.0	0.0
3-D LINE ELEMENT	4	SIZE	0.250	LENGTH	3.000	AREA	0.530
STRESS AT NODES 105,106			5.5,		5.3	ALLOWABLE	20.0
STRESS EXPANSION FOR NODE 105							
0.5	-0.6		4.4		-0.4	0.0	0.0
STRESS EXPANSION FOR NODE 106							
0.5	-0.6		4.4		-0.6	0.0	0.0

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\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

6	3-D LINE ELEMENT	5	SIZE	0.250	LENGTH	3.000	AREA	0.530	
7	STRESS AT NODES 107,108			3.3,	3.5	ALLOWABLE		20.0	
8	STRESS EXPANSION FOR NODE 107								
9	0.3	-0.6	2.3		-0.6		0.0	0.0	
10	STRESS EXPANSION FOR NODE 108								
11	0.3	-0.6	2.3		-0.4		0.0	0.0	
12									
13	3-D LINE ELEMENT	6	SIZE	0.250	LENGTH	3.000	AREA	0.530	
14	STRESS AT NODES 109,110			3.8,	4.0	ALLOWABLE		20.0	
15	STRESS EXPANSION FOR NODE 109								
16	0.3	-0.6	2.3		-0.1		0.0	0.0	
17	STRESS EXPANSION FOR NODE 110								
18	0.3	-0.6	2.3		0.1		0.0	0.0	
19									
20	3-D LINE ELEMENT	7	SIZE	0.250	LENGTH	3.000	AREA	0.530	
21	STRESS AT NODES 111,112			4.3,	4.5	ALLOWABLE		20.0	
22	STRESS EXPANSION FOR NODE 111								
23	0.3	-0.6	2.3		0.4		0.0	0.0	
24	STRESS EXPANSION FOR NODE 112								
25	0.3	-0.6	2.3		0.6		0.0	0.0	
26									
27	3-D LINE ELEMENT	8	SIZE	0.250	LENGTH	3.000	AREA	0.530	
28	STRESS AT NODES 115,116			6.0,	5.8	ALLOWABLE		20.0	
29	STRESS EXPANSION FOR NODE 115								
30	0.5	0.6	4.4		0.1		0.0	0.0	
31	STRESS EXPANSION FOR NODE 116								
32	0.5	0.6	4.4		-0.1		0.0	0.0	
33									
34	3-D LINE ELEMENT	9	SIZE	0.250	LENGTH	3.000	AREA	0.530	
35	STRESS AT NODES 117,118			5.5,	5.3	ALLOWABLE		20.0	
36	STRESS EXPANSION FOR NODE 117								
37	0.5	0.6	4.4		-0.4		0.0	0.0	
38	STRESS EXPANSION FOR NODE 118								
39	0.5	0.6	4.4		-0.6		0.0	0.0	
40									
41	3-D LINE ELEMENT	10	SIZE	0.250	LENGTH	3.000	AREA	0.530	
42	STRESS AT NODES 119,120			3.3,	3.5	ALLOWABLE		20.0	
43	STRESS EXPANSION FOR NODE 119								
44	0.3	0.6	2.3		-0.6		0.0	0.0	
45	STRESS EXPANSION FOR NODE 120								
46	0.3	0.6	2.3		-0.4		0.0	0.0	
47									
48	3-D LINE ELEMENT	11	SIZE	0.250	LENGTH	3.000	AREA	0.530	
49	STRESS AT NODES 121,122			3.8,	4.0	ALLOWABLE		20.0	
50	STRESS EXPANSION FOR NODE 121								
51	0.3	0.6	2.3		-0.1		0.0	0.0	
52	STRESS EXPANSION FOR NODE 122								
53	0.3	0.6	2.3		0.1		0.0	0.0	
54									
55	3-D LINE ELEMENT	12	SIZE	0.250	LENGTH	3.000	AREA	0.530	
56	STRESS AT NODES 123,124			4.3,	4.5	ALLOWABLE		20.0	
57	STRESS EXPANSION FOR NODE 123								
	0.3	0.6	2.3		0.4		0.0	0.0	
	STRESS EXPANSION FOR NODE 124								
	0.3	0.6	2.3		0.6		0.0	0.0	

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\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

6	3-D LINE ELEMENT	13	SIZE	0.250	LENGTH	3.000	AREA	0.530
7	STRESS AT NODES 125,126			3.1,	3.2	ALLOWABLE		20.0
8	STRESS EXPANSION FOR NODE 125							
9	-0.6	-0.6		-4.8	0.6		0.0	0.0
10	STRESS EXPANSION FOR NODE 126							
11	-0.6	-0.6		-4.8	0.4		0.0	0.0
12								
13	3-D LINE ELEMENT	14	SIZE	0.250	LENGTH	3.000	AREA	0.530
14	STRESS AT NODES 127,128			3.5,	3.7	ALLOWABLE		20.0
15	STRESS EXPANSION FOR NODE 127							
16	-0.6	-0.6		-4.8	0.1		0.0	0.0
17	STRESS EXPANSION FOR NODE 128							
18	-0.6	-0.6		-4.8	-0.1		0.0	0.0
19								
20	3-D LINE ELEMENT	15	SIZE	0.250	LENGTH	3.000	AREA	0.530
21	STRESS AT NODES 129,130			4.0,	4.2	ALLOWABLE		20.0
22	STRESS EXPANSION FOR NODE 129							
23	-0.6	-0.6		-4.8	-0.4		0.0	0.0
24	STRESS EXPANSION FOR NODE 130							
25	-0.6	-0.6		-4.8	-0.6		0.0	0.0
26								
27	3-D LINE ELEMENT	16	SIZE	0.250	LENGTH	3.000	AREA	0.530
28	STRESS AT NODES 131,132			4.5,	4.6	ALLOWABLE		20.0
29	STRESS EXPANSION FOR NODE 131							
30	-0.6	-0.6		-4.8	-0.9		0.0	0.0
31	STRESS EXPANSION FOR NODE 132							
32	-0.6	-0.6		-4.8	-1.1		0.0	0.0
33								
34	3-D LINE ELEMENT	17	SIZE	0.250	LENGTH	3.000	AREA	0.530
35	STRESS AT NODES 133,134			4.9,	5.1	ALLOWABLE		20.0
36	STRESS EXPANSION FOR NODE 133							
37	-0.6	-0.6		-4.8	-1.4		0.0	0.0
38	STRESS EXPANSION FOR NODE 134							
39	-0.6	-0.6		-4.8	-1.6		0.0	0.0
40								
41	3-D LINE ELEMENT	18	SIZE	0.250	LENGTH	3.000	AREA	0.530
42	STRESS AT NODES 135,136			5.4,	5.6	ALLOWABLE		20.0
43	STRESS EXPANSION FOR NODE 135							
44	-0.6	-0.6		-4.8	-1.9		0.0	0.0
45	STRESS EXPANSION FOR NODE 136							
46	-0.6	-0.6		-4.8	-2.1		0.0	0.0
47								
48	3-D LINE ELEMENT	19	SIZE	0.250	LENGTH	3.000	AREA	0.530
49	STRESS AT NODES 137,138			7.6,	7.5	ALLOWABLE		20.0
50	STRESS EXPANSION FOR NODE 137							
51	-0.8	-0.6		-6.8	-2.1		0.0	0.0
52	STRESS EXPANSION FOR NODE 138							
53	-0.8	-0.6		-6.8	-1.9		0.0	0.0
54								
55	3-D LINE ELEMENT	20	SIZE	0.250	LENGTH	3.000	AREA	0.530
56	STRESS AT NODES 139,140			7.1,	7.0	ALLOWABLE		20.0
57	STRESS EXPANSION FOR NODE 139							
58	-0.8	-0.6		-6.8	-1.6		0.0	0.0
59	STRESS EXPANSION FOR NODE 140							
60	-0.8	-0.6		-6.8	-1.4		0.0	0.0

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\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

3-D LINE ELEMENT 21	SIZE 0.250	LENGTH 3.000	AREA 0.530
STRESS AT NODES 141,142		6.7,	6.5 ALLOWABLE 20.0
STRESS EXPANSION FOR NODE 141			
-0.8	-0.6	-6.8	-1.1 0.0 0.0
STRESS EXPANSION FOR NODE 142			
-0.8	-0.6	-6.8	-0.9 0.0 0.0
3-D LINE ELEMENT 22	SIZE 0.250	LENGTH 3.000	AREA 0.530
STRESS AT NODES 143,144		2.7,	2.9 ALLOWABLE 20.0
STRESS EXPANSION FOR NODE 143			
-0.6	0.6	-4.8	0.6 0.0 0.0
STRESS EXPANSION FOR NODE 144			
-0.6	0.6	-4.8	0.4 0.0 0.0
3-D LINE ELEMENT 23	SIZE 0.250	LENGTH 3.000	AREA 0.530
STRESS AT NODES 145,146		3.2,	3.4 ALLOWABLE 20.0
STRESS EXPANSION FOR NODE 145			
-0.6	0.6	-4.8	0.1 0.0 0.0
STRESS EXPANSION FOR NODE 146			
-0.6	0.6	-4.8	-0.1 0.0 0.0
3-D LINE ELEMENT 24	SIZE 0.250	LENGTH 3.000	AREA 0.530
STRESS AT NODES 147,148		3.7,	3.9 ALLOWABLE 20.0
STRESS EXPANSION FOR NODE 147			
-0.6	0.6	-4.8	-0.4 0.0 0.0
STRESS EXPANSION FOR NODE 148			
-0.6	0.6	-4.8	-0.6 0.0 0.0
3-D LINE ELEMENT 25	SIZE 0.250	LENGTH 3.000	AREA 0.530
STRESS AT NODES 149,150		4.2,	4.4 ALLOWABLE 20.0
STRESS EXPANSION FOR NODE 149			
-0.6	0.6	-4.8	-0.9 0.0 0.0
STRESS EXPANSION FOR NODE 150			
-0.6	0.6	-4.8	-1.1 0.0 0.0
3-D LINE ELEMENT 26	SIZE 0.250	LENGTH 3.000	AREA 0.530
STRESS AT NODES 151,152		4.7,	4.9 ALLOWABLE 20.0
STRESS EXPANSION FOR NODE 151			
-0.6	0.6	-4.8	-1.4 0.0 0.0
STRESS EXPANSION FOR NODE 152			
-0.6	0.6	-4.8	-1.6 0.0 0.0
3-D LINE ELEMENT 27	SIZE 0.250	LENGTH 3.000	AREA 0.530
STRESS AT NODES 153,154		5.2,	5.4 ALLOWABLE 20.0
STRESS EXPANSION FOR NODE 153			
-0.6	0.6	-4.8	-1.9 0.0 0.0
STRESS EXPANSION FOR NODE 154			
-0.6	0.6	-4.8	-2.1 0.0 0.0
3-D LINE ELEMENT 28	SIZE 0.250	LENGTH 3.000	AREA 0.530
STRESS AT NODES 155,156		7.5,	7.3 ALLOWABLE 20.0
STRESS EXPANSION FOR NODE 155			
-0.8	0.6	-6.8	-2.1 0.0 0.0
STRESS EXPANSION FOR NODE 156			
-0.8	0.6	-6.8	-1.9 0.0 0.0

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MJM 8-27-87

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\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

6	3-D LINE ELEMENT	29	SIZE	0.250	LENGTH	3.000	AREA	0.530
7	STRESS AT NODES 157,158			7.0,	6.8	ALLOWABLE		20.0
8	STRESS EXPANSION FOR NODE 157							
9	-0.8	0.6		-6.8	-1.6		0.0	0.0
10	STRESS EXPANSION FOR NODE 158							
11	-0.8	0.6		-6.8	-1.4		0.0	0.0
12								
13	3-D LINE ELEMENT	30	SIZE	0.250	LENGTH	3.000	AREA	0.530
14	STRESS AT NODES 159,160			6.5,	6.3	ALLOWABLE		20.0
15	STRESS EXPANSION FOR NODE 159							
16	-0.8	0.6		-6.8	-1.1		0.0	0.0
17	STRESS EXPANSION FOR NODE 160							
18	-0.8	0.6		-6.8	-0.9		0.0	0.0
19								
20	3-D LINE ELEMENT	31	SIZE	0.250	LENGTH	6.000	AREA	1.061
21	STRESS AT NODES 106,107			5.3,	3.3	ALLOWABLE		20.0
22	STRESS EXPANSION FOR NODE 106							
23	0.5	-0.6		4.4	-0.6		0.0	0.0
24	STRESS EXPANSION FOR NODE 107							
25	0.3	-0.6		2.3	-0.6		0.0	0.0
26								
27	3-D LINE ELEMENT	32	SIZE	0.250	LENGTH	6.000	AREA	1.061
28	STRESS AT NODES 118,119			5.3,	3.3	ALLOWABLE		20.0
29	STRESS EXPANSION FOR NODE 118							
30	0.5	0.6		4.4	-0.6		0.0	0.0
31	STRESS EXPANSION FOR NODE 119							
32	0.3	0.6		2.3	-0.6		0.0	0.0
33								
34	3-D LINE ELEMENT	33	SIZE	0.250	LENGTH	5.750	AREA	1.016
35	STRESS AT NODES 101,165			7.8,	5.8	ALLOWABLE		20.0
36	STRESS EXPANSION FOR NODE 101							
37	0.5	-0.6		4.4	1.9		0.0	0.0
38	STRESS EXPANSION FOR NODE 165							
39	0.3	-0.6		2.4	1.9		0.0	0.0
40								
41	3-D LINE ELEMENT	34	SIZE	0.250	LENGTH	5.750	AREA	1.016
42	STRESS AT NODES 113,166			7.8,	5.8	ALLOWABLE		20.0
43	STRESS EXPANSION FOR NODE 113							
44	0.5	0.6		4.4	1.9		0.0	0.0
45	STRESS EXPANSION FOR NODE 166							
46	0.3	0.6		2.4	1.9		0.0	0.0
47								
48	3-D LINE ELEMENT	35	SIZE	0.250	LENGTH	5.500	AREA	0.972
49	STRESS AT NODES 125,167			3.1,	4.9	ALLOWABLE		20.0
50	STRESS EXPANSION FOR NODE 125							
51	-0.6	-0.6		-4.8	0.6		0.0	0.0
52	STRESS EXPANSION FOR NODE 167							
53	-0.8	-0.6		-6.6	0.6		0.0	0.0
54								
55	3-D LINE ELEMENT	36	SIZE	0.250	LENGTH	5.500	AREA	0.972
56	STRESS AT NODES 143,168			2.7,	4.6	ALLOWABLE		20.0
57	STRESS EXPANSION FOR NODE 143							
	-0.6	0.6		-4.8	0.6		0.0	0.0
	STRESS EXPANSION FOR NODE 168							
	-0.8	0.6		-6.6	0.6		0.0	0.0

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\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

3-D LINE ELEMENT	37	SIZE	0.250	LENGTH	6.000	AREA	1.061
STRESS AT NODES 161,162			7.9,	5.8	ALLOWABLE	20.0	
STRESS EXPANSION FOR NODE 161							
	-0.8	-0.6	-6.8	-2.4	0.0	0.0	
STRESS EXPANSION FOR NODE 162							
	-0.6	-0.6	-4.8	-2.4	0.0	0.0	

3-D LINE ELEMENT	38.	SIZE	0.250	LENGTH	6.000	AREA	1.061
STRESS AT NODES 163,164			7.7,	5.7	ALLOWABLE	20.0	
STRESS EXPANSION FOR NODE 163							
	-0.8	0.6	-6.8	-2.4	0.0	0.0	
STRESS EXPANSION FOR NODE 164							
	-0.6	0.6	-4.8	-2.4	0.0	0.0	

FORCE DEFINITION	NODE	DIRECT	ELEMENT	39	NODE	200
FX	-11.80	FY	49.80	FZ	0.00	
MX	-3267.00	MY	0.00	MZ	0.00	

MAXIMUM ABSOLUTE STRESS ON ELEMENT 37 = 7.9 KSI

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* OBE

\*\*\*\*\* REPORT SUMMARY, FOR A 4 LOAD STEP ANALYSIS \*\*\*\*\*

WORST CASE ANALYSIS.....NO

ELEMENT STRESSES PRINTED....YES, STRESS SUMMATION MODE WAS DIRECT

ELEMENT STATISTICS

TYPE	DESCRIPTION	NUMBER USED
1	3-DIMENSIONAL LINE ELEMENT.....	36
2	3-DIMENSIONAL LINE ELEMENT.....	2
3	FORCE DEFINITION NODE - DIRECT.....	1

TOTAL NUMBER OF ELEMENTS = 39, NODES = 69

SYSTEM PROPERTIES

LOAD STEP..... 3 DIRECTION, LOCATION OR AXIS

	X	Y	Z
CENTROIDS, X AXIS.....		0.00	22.50
Y AXIS.....	0.00		22.50
Z AXIS.....	-19.99	-0.00	
SHEAR AREAS.....	33.23	33.23	0.00
POLAR MOMENTS OF INERTIA..	6299.12	6299.12	15729.29
TRANSLATED FORCES.....	-11.80	49.80	0.00
TRANSLATED MOMENTS.....	-2146.55	265.49	995.28

MAXIMUM ELEMENT STRESSES

DESCRIPTION	NUMBER USED	ELEMENT	NODE	STRESS	ALLOWABLE	COMPAR FACTO
3-D LINE ELEMENT	36	37	161	7.9 KSI	20.0 KSI	2.53
3-D LINE ELEMENT	2	2	113	7.8 KSI	20.0 KSI	2.56
3-D LINE ELEMENT	36	37	161	7.9 KSI	20.0 KSI	2.53
3-D LINE ELEMENT	2	2	113	7.8 KSI	20.0 KSI	2.56

\*\*\*\*\* JOB COMPLETED \*\*\*\*\*

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	V	V	V	V	V	V	V	V	V	V	V	V	V	V	V	V
1.	AEP * 150T CRANE * 50T LOAD * TRUCK TO GIRDER * WELDS * SSE															
2.	79604	ASZ	4.	1.	1.	0.				1.	1.					
3.	1.	1.														
4.	2.	1.														
5.	3.	4.														
6.	-1.															
7.	0.25	27.3				1.				3.						
8.	0.3125	27.3				1.				3.						
9.	-1.															
10.	101.	102.		2.	2.			2.		12.						
11.	103.	104.		1.	1.			5.		2.						
12.	115.	116.		1.	1.			23.		2.						
13.	106.	107.		1.	1.			2.		12.						
14.	101.	165.		1.	1.											
15.	113.	166.		1.	1.											
16.	125.	167.		1.	1.											
17.	143.	168.		1.	1.											
18.	161.	162.		1.	1.			2.		2.						
19.	200.			3.												
20.	-1.															
21.	-12.	12.						-17.625								
22.	101.			9.5				8.8125		35.375						
23.	102.			-13.5				8.8125		35.375						
24.	103.			-18.5				8.8125		35.375						
25.	104.			-21.5				8.8125		35.375						
26.	105.			-26.5				8.8125		35.375						
27.	106.			-29.5				8.8125		35.375						
28.	107.			-29.5				8.8125		29.375						
29.	108.			-26.5				8.8125		29.375						
30.	109.			-21.5				8.8125		29.375						
31.	110.			-18.5				8.8125		29.375						
32.	111.			-13.5				8.8125		29.375						
33.	112.			-10.5				8.8125		29.375						
34.	-18.	18.						-17.625								
35.	125.			-10.5				8.8125		8.5						
36.	126.			-13.5				8.8125		8.5						
37.	127.			-18.5				8.8125		8.5						
38.	128.			-21.5				8.8125		8.5						
39.	129.			-26.5				8.8125		8.5						
40.	130.			-29.5				8.8125		8.5						
41.	131.			-34.5				8.8125		8.5						
42.	132.			-37.5				8.8125		8.5						
43.	133.			-42.5				8.8125		8.5						
44.	134.			-45.5				8.8125		8.5						
45.	135.			-50.5				8.8125		8.5						
46.	136.			-53.5				8.8125		8.5						
47.	137.			-53.5				8.8125		2.5						
48.	138.			-50.5				8.8125		2.5						
49.	139.			-45.5				8.8125		2.5						
50.	140.			-42.5				8.8125		2.5						
51.	141.			-37.5				8.8125		2.5						
52.	142.			-34.5				8.8125		2.5						
53.	-2.	2.						-17.625								
54.	161.			-57.5				8.8125		2.5						
55.	162.			-57.5				8.8125		8.5						
56.	165.			9.5				8.8125		29.625						
57.	166.			9.5				-8.8125		29.625						
58.	167.			-10.5				8.8125		3.0						
59.	168.			-10.5				-8.8125		3.0						
60.	200.			0.				0.		0.						
61.	9799.															
62.	101.	1.	168.													
63.	-1.															
64.	200.	22.2	78.8		0.					-5384.						
65.	-1.															
66.	200.	22.2	-78.8		0.					5384.						
67.	-1.															
68.	200.	-22.2	78.8		0.					-5384.						
69.	-1.															
70.	200.	-22.2	-78.8		0.					5384.						
71.	-1.															
72.	FINISH															
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PAGE 4-62 OF 112

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MJM 8-27-87

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* SSE

\*\*\*\*\* WORST CASE ANALYSIS SUMMARY \*\*\*\*\*

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* SSE

LOAD STEP.....	1	DIRECTION		
		X	Y	Z
TRANSLATED FORCES.....	22.20	78.80	0.00	
TRANSLATED MOMENTS.....	-3611.08	-499.48	1574.85	
NUMBER OF FORCE DEFINITION NODES...	1			
MAXIMUM ABSOLUTE STRESS ON ELEMENT 38 =	13.2 KSI			
COMPARISON FACTOR MATCH ON ELEMENT 38 =	2.0639749			

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* SSE

LOAD STEP.....	2	DIRECTION		
		X	Y	Z
TRANSLATED FORCES.....	22.20	-78.80	0.00	
TRANSLATED MOMENTS.....	3611.08	-499.48	-1574.85	
NUMBER OF FORCE DEFINITION NODES...	1			
MAXIMUM ABSOLUTE STRESS ON ELEMENT 37 =	13.2 KSI			
COMPARISON FACTOR MATCH ON ELEMENT 37 =	2.0639758			

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* SSE

LOAD STEP.....	3	DIRECTION		
		X	Y	Z
TRANSLATED FORCES.....	-22.20	78.80	0.00	
TRANSLATED MOMENTS.....	-3611.08	499.48	1574.85	
NUMBER OF FORCE DEFINITION NODES...	1			
MAXIMUM ABSOLUTE STRESS ON ELEMENT 37 =	13.2 KSI			
COMPARISON FACTOR MATCH ON ELEMENT 37 =	2.0639749			

AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* SSE

LOAD STEP.....	4	DIRECTION		
		X	Y	Z
TRANSLATED FORCES.....	-22.20	-78.80	0.00	
TRANSLATED MOMENTS.....	3611.08	499.48	-1574.85	
NUMBER OF FORCE DEFINITION NODES...	1			
MAXIMUM ABSOLUTE STRESS ON ELEMENT 38 =	13.2 KSI			
COMPARISON FACTOR MATCH ON ELEMENT 38 =	2.0639758			

\*\*\*\*\* WORST CASE ANALYSIS COMPLETE \*\*\*\*\*

WORST CASE OCCURED DURING LOAD STEP 1

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* SSE

\*\*\*\*\* SYSTEM PROPERTIES \*\*\*\*\*

LOAD STEP..... 1	DIRECTION, LOCATION OR AXIS		
	X	Y	Z
CENTROIDS, X AXIS.....		0.00	22.50
Y AXIS.....	0.00		22.50
Z AXIS.....	-19.99	-0.00	
SHEAR AREAS.....	33.23	33.23	0.00
POLAR MOMENTS OF INERTIA..	6299.12	6299.12	15729.29
TRANSLATED FORCES.....	22.20	78.80	0.00
TRANSLATED MOMENTS.....	-3611.08	-499.48	1574.85
NUMBER OF FORCE DEFINITION NODES... 1			

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

DIRECT STRESSES ARE EQUALLY DISTRIBUTED THROUGHOUT SHEAR AREA

DRX 0.7 DRY 2.4 DRZ 0.0

3-D LINE ELEMENT 1	SIZE 0.313	LENGTH 23.000	AREA 5.082
STRESS AT NODES 101,102	12.8,	10.5	ALLOWABLE 27.3
STRESS EXPANSION FOR NODE 101			
-1.0	-0.9	7.4	3.0 0.0 0.0
STRESS EXPANSION FOR NODE 102			
-1.0	-0.9	7.4	0.6 0.0 0.0
3-D LINE ELEMENT 2	SIZE 0.313	LENGTH 23.000	AREA 5.082
STRESS AT NODES 113,114	12.7,	10.4	ALLOWABLE 27.3
STRESS EXPANSION FOR NODE 113			
-1.0	0.9	7.4	3.0 0.0 0.0
STRESS EXPANSION FOR NODE 114			
-1.0	0.9	7.4	0.6 0.0 0.0
3-D LINE ELEMENT 3	SIZE 0.250	LENGTH 3.000	AREA 0.530
STRESS AT NODES 103,104	10.0,	9.7	ALLOWABLE 27.3
STRESS EXPANSION FOR NODE 103			
-1.0	-0.9	7.4	0.1 0.0 0.0
STRESS EXPANSION FOR NODE 104			
-1.0	-0.9	7.4	-0.2 0.0 0.0
3-D LINE ELEMENT 4	SIZE 0.250	LENGTH 3.000	AREA 0.530
STRESS AT NODES 105,106	9.2,	8.9	ALLOWABLE 27.3
STRESS EXPANSION FOR NODE 105			
-1.0	-0.9	7.4	-0.7 0.0 0.0
STRESS EXPANSION FOR NODE 106			
-1.0	-0.9	7.4	-1.0 0.0 0.0

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* SSE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

6	3-D LINE ELEMENT	5	SIZE	0.250	LENGTH	3.000	AREA	0.530
7	STRESS AT NODES 107,108			5.4,	5.7	ALLOWABLE		27.3
8	STRESS EXPANSION FOR NODE 107							
9	-0.5	-0.9	3.9		-1.0		0.0	0.0
10	STRESS EXPANSION FOR NODE 108							
11	-0.5	-0.9	3.9		-0.7		0.0	0.0
13	3-D LINE ELEMENT	6	SIZE	0.250	LENGTH	3.000	AREA	0.530
14	STRESS AT NODES 109,110			6.2,	6.5	ALLOWABLE		27.3
15	STRESS EXPANSION FOR NODE 109							
16	-0.5	-0.9	3.9		-0.2		0.0	0.0
17	STRESS EXPANSION FOR NODE 110							
18	-0.5	-0.9	3.9		0.1		0.0	0.0
20	3-D LINE ELEMENT	7	SIZE	0.250	LENGTH	3.000	AREA	0.530
21	STRESS AT NODES 111,112			7.0,	7.3	ALLOWABLE		27.3
22	STRESS EXPANSION FOR NODE 111							
23	-0.5	-0.9	3.9		0.6		0.0	0.0
24	STRESS EXPANSION FOR NODE 112							
25	-0.5	-0.9	3.9		0.9		0.0	0.0
27	3-D LINE ELEMENT	8	SIZE	0.250	LENGTH	3.000	AREA	0.530
28	STRESS AT NODES 115,116			9.9,	9.6	ALLOWABLE		27.3
29	STRESS EXPANSION FOR NODE 115							
30	-1.0	0.9	7.4		0.1		0.0	0.0
31	STRESS EXPANSION FOR NODE 116							
32	-1.0	0.9	7.4		-0.2		0.0	0.0
34	3-D LINE ELEMENT	9	SIZE	0.250	LENGTH	3.000	AREA	0.530
35	STRESS AT NODES 117,118			9.1,	8.8	ALLOWABLE		27.3
36	STRESS EXPANSION FOR NODE 117							
37	-1.0	0.9	7.4		-0.7		0.0	0.0
38	STRESS EXPANSION FOR NODE 118							
39	-1.0	0.9	7.4		-1.0		0.0	0.0
41	3-D LINE ELEMENT	10	SIZE	0.250	LENGTH	3.000	AREA	0.530
42	STRESS AT NODES 119,120			5.5,	5.7	ALLOWABLE		27.3
43	STRESS EXPANSION FOR NODE 119							
44	-0.5	0.9	3.9		-1.0		0.0	0.0
45	STRESS EXPANSION FOR NODE 120							
46	-0.5	0.9	3.9		-0.7		0.0	0.0
48	3-D LINE ELEMENT	11	SIZE	0.250	LENGTH	3.000	AREA	0.530
49	STRESS AT NODES 121,122			6.2,	6.5	ALLOWABLE		27.3
50	STRESS EXPANSION FOR NODE 121							
51	-0.5	0.9	3.9		-0.2		0.0	0.0
52	STRESS EXPANSION FOR NODE 122							
53	-0.5	0.9	3.9		0.1		0.0	0.0
55	3-D LINE ELEMENT	12	SIZE	0.250	LENGTH	3.000	AREA	0.530
56	STRESS AT NODES 123,124			7.0,	7.3	ALLOWABLE		27.3
57	STRESS EXPANSION FOR NODE 123							
	-0.5	0.9	3.9		0.6		0.0	0.0
	STRESS EXPANSION FOR NODE 124							
	-0.5	0.9	3.9		0.9		0.0	0.0

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* SSE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

6	3-D LINE ELEMENT	13	SIZE	0.250	LENGTH	3.000	AREA	0.530
7	STRESS AT NODES 125, 126		4.8,	5.1	ALLOWABLE	27.3		
8	STRESS EXPANSION FOR NODE 125							
9	1.1	-0.9	-8.0	0.9	0.0	0.0		
10	STRESS EXPANSION FOR NODE 126							
11	1.1	-0.9	-8.0	0.6	0.0	0.0		
12								
13	3-D LINE ELEMENT	14	SIZE	0.250	LENGTH	3.000	AREA	0.530
14	STRESS AT NODES 127, 128		5.6,	5.9	ALLOWABLE	27.3		
15	STRESS EXPANSION FOR NODE 127							
16	1.1	-0.9	-8.0	0.1	0.0	0.0		
17	STRESS EXPANSION FOR NODE 128							
18	1.1	-0.9	-8.0	-0.2	0.0	0.0		
19								
20	3-D LINE ELEMENT	15	SIZE	0.250	LENGTH	3.000	AREA	0.530
21	STRESS AT NODES 129, 130		6.4,	6.7	ALLOWABLE	27.3		
22	STRESS EXPANSION FOR NODE 129							
23	1.1	-0.9	-8.0	-0.7	0.0	0.0		
24	STRESS EXPANSION FOR NODE 130							
25	1.1	-0.9	-8.0	-1.0	0.0	0.0		
26								
27	3-D LINE ELEMENT	16	SIZE	0.250	LENGTH	3.000	AREA	0.530
28	STRESS AT NODES 131, 132		7.2,	7.5	ALLOWABLE	27.3		
29	STRESS EXPANSION FOR NODE 131							
30	1.1	-0.9	-8.0	-1.5	0.0	0.0		
31	STRESS EXPANSION FOR NODE 132							
32	1.1	-0.9	-8.0	-1.8	0.0	0.0		
33								
34	3-D LINE ELEMENT	17	SIZE	0.250	LENGTH	3.000	AREA	0.530
35	STRESS AT NODES 133, 134		8.0,	8.3	ALLOWABLE	27.3		
36	STRESS EXPANSION FOR NODE 133							
37	1.1	-0.9	-8.0	-2.3	0.0	0.0		
38	STRESS EXPANSION FOR NODE 134							
39	1.1	-0.9	-8.0	-2.6	0.0	0.0		
40								
41	3-D LINE ELEMENT	18	SIZE	0.250	LENGTH	3.000	AREA	0.530
42	STRESS AT NODES 135, 136		8.8,	9.1	ALLOWABLE	27.3		
43	STRESS EXPANSION FOR NODE 135							
44	1.1	-0.9	-8.0	-3.1	0.0	0.0		
45	STRESS EXPANSION FOR NODE 136							
46	1.1	-0.9	-8.0	-3.4	0.0	0.0		
47								
48	3-D LINE ELEMENT	19	SIZE	0.250	LENGTH	3.000	AREA	0.530
49	STRESS AT NODES 137, 138		12.5,	12.2	ALLOWABLE	27.3		
50	STRESS EXPANSION FOR NODE 137							
51	1.6	-0.9	-11.5	-3.4	0.0	0.0		
52	STRESS EXPANSION FOR NODE 138							
53	1.6	-0.9	-11.5	-3.1	0.0	0.0		
54								
55	3-D LINE ELEMENT	20	SIZE	0.250	LENGTH	3.000	AREA	0.530
56	STRESS AT NODES 139, 140		11.7,	11.4	ALLOWABLE	27.3		
57	STRESS EXPANSION FOR NODE 139							
	1.6	-0.9	-11.5	-2.6	0.0	0.0		
	STRESS EXPANSION FOR NODE 140							
	1.6	-0.9	-11.5	-2.3	0.0	0.0		

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* SSE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

6	3-D LINE ELEMENT	21	SIZE	0.250	LENGTH	3.000	AREA	0.530
7	STRESS AT NODES 141,142			10.9,	10.6	ALLOWABLE		27.3
8	STRESS EXPANSION FOR NODE 141							
9	1.6	-0.9	-11.5		-1.8		0.0	0.0
10	STRESS EXPANSION FOR NODE 142							
11	1.6	-0.9	-11.5		-1.5		0.0	0.0
12	-----							
13	3-D LINE ELEMENT	22	SIZE	0.250	LENGTH	3.000	AREA	0.530
14	STRESS AT NODES 143,144			5.4,	5.7	ALLOWABLE		27.3
15	STRESS EXPANSION FOR NODE 143							
16	1.1	0.9	-8.0		0.9		0.0	0.0
17	STRESS EXPANSION FOR NODE 144							
18	1.1	0.9	-8.0		0.6		0.0	0.0
19	-----							
20	3-D LINE ELEMENT	23	SIZE	0.250	LENGTH	3.000	AREA	0.530
21	STRESS AT NODES 145,146			6.1,	6.4	ALLOWABLE		27.3
22	STRESS EXPANSION FOR NODE 145							
23	1.1	0.9	-8.0		0.1		0.0	0.0
24	STRESS EXPANSION FOR NODE 146							
25	1.1	0.9	-8.0		-0.2		0.0	0.0
26	-----							
27	3-D LINE ELEMENT	24	SIZE	0.250	LENGTH	3.000	AREA	0.530
28	STRESS AT NODES 147,148			6.8,	7.1	ALLOWABLE		27.3
29	STRESS EXPANSION FOR NODE 147							
30	1.1	0.9	-8.0		-0.7		0.0	0.0
31	STRESS EXPANSION FOR NODE 148							
32	1.1	0.9	-8.0		-1.0		0.0	0.0
33	-----							
34	3-D LINE ELEMENT	25	SIZE	0.250	LENGTH	3.000	AREA	0.530
35	STRESS AT NODES 149,150			7.6,	7.9	ALLOWABLE		27.3
36	STRESS EXPANSION FOR NODE 149							
37	1.1	0.9	-8.0		-1.5		0.0	0.0
38	STRESS EXPANSION FOR NODE 150							
39	1.1	0.9	-8.0		-1.8		0.0	0.0
40	-----							
41	3-D LINE ELEMENT	26	SIZE	0.250	LENGTH	3.000	AREA	0.530
42	STRESS AT NODES 151,152			8.3,	8.6	ALLOWABLE		27.3
43	STRESS EXPANSION FOR NODE 151							
44	1.1	0.9	-8.0		-2.3		0.0	0.0
45	STRESS EXPANSION FOR NODE 152							
46	1.1	0.9	-8.0		-2.6		0.0	0.0
47	-----							
48	3-D LINE ELEMENT	27	SIZE	0.250	LENGTH	3.000	AREA	0.530
49	STRESS AT NODES 153,154			9.1,	9.4	ALLOWABLE		27.3
50	STRESS EXPANSION FOR NODE 153							
51	1.1	0.9	-8.0		-3.1		0.0	0.0
52	STRESS EXPANSION FOR NODE 154							
53	1.1	0.9	-8.0		-3.4		0.0	0.0
54	-----							
55	3-D LINE ELEMENT	28	SIZE	0.250	LENGTH	3.000	AREA	0.530
56	STRESS AT NODES 155,156			12.8,	12.5	ALLOWABLE		27.3
57	STRESS EXPANSION FOR NODE 155							
	1.6	0.9	-11.5		-3.4		0.0	0.0
	STRESS EXPANSION FOR NODE 156							
	1.6	0.9	-11.5		-3.1		0.0	0.0

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* SSE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

6	3-D LINE ELEMENT	29	SIZE	0.250	LENGTH	3.000	AREA	0.530
7	STRESS AT NODES 157,158			12.1,	11.8	ALLOWABLE		27.3
8	STRESS EXPANSION FOR NODE 157							
9	1.6	0.9		-11.5	-2.6		0.0	0.0
10	STRESS EXPANSION FOR NODE 158							
11	1.6	0.9		-11.5	-2.3		0.0	0.0
12								
13	3-D LINE ELEMENT	30	SIZE	0.250	LENGTH	3.000	AREA	0.530
14	STRESS AT NODES 159,160			11.3,	11.0	ALLOWABLE		27.3
15	STRESS EXPANSION FOR NODE 159							
16	1.6	0.9		-11.5	-1.8		0.0	0.0
17	STRESS EXPANSION FOR NODE 160							
18	1.6	0.9		-11.5	-1.5		0.0	0.0
19								
20	3-D LINE ELEMENT	31	SIZE	0.250	LENGTH	6.000	AREA	1.061
21	STRESS AT NODES 106,107			8.9,	5.4	ALLOWABLE		27.3
22	STRESS EXPANSION FOR NODE 106							
23	-1.0	-0.9		7.4	-1.0		0.0	0.0
24	STRESS EXPANSION FOR NODE 107							
25	-0.5	-0.9		3.9	-1.0		0.0	0.0
26								
27	3-D LINE ELEMENT	32	SIZE	0.250	LENGTH	6.000	AREA	1.061
28	STRESS AT NODES 118,119			8.8,	5.5	ALLOWABLE		27.3
29	STRESS EXPANSION FOR NODE 118							
30	-1.0	0.9		7.4	-1.0		0.0	0.0
31	STRESS EXPANSION FOR NODE 119							
32	-0.5	0.9		3.9	-1.0		0.0	0.0
33								
34	3-D LINE ELEMENT	33	SIZE	0.250	LENGTH	5.750	AREA	1.016
35	STRESS AT NODES 101,165			12.8,	9.4	ALLOWABLE		27.3
36	STRESS EXPANSION FOR NODE 101							
37	-1.0	-0.9		7.4	3.0		0.0	0.0
38	STRESS EXPANSION FOR NODE 165							
39	-0.6	-0.9		4.1	3.0		0.0	0.0
40								
41	3-D LINE ELEMENT	34	SIZE	0.250	LENGTH	5.750	AREA	1.016
42	STRESS AT NODES 113,166			12.7,	9.5	ALLOWABLE		27.3
43	STRESS EXPANSION FOR NODE 113							
44	-1.0	0.9		7.4	3.0		0.0	0.0
45	STRESS EXPANSION FOR NODE 166							
46	-0.6	0.9		4.1	3.0		0.0	0.0
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48	3-D LINE ELEMENT	35	SIZE	0.250	LENGTH	5.500	AREA	0.972
49	STRESS AT NODES 125,167			4.8,	8.0	ALLOWABLE		27.3
50	STRESS EXPANSION FOR NODE 125							
51	1.1	-0.9		-8.0	0.9		0.0	0.0
52	STRESS EXPANSION FOR NODE 167							
53	1.5	-0.9		-11.2	0.9		0.0	0.0
54								
55	3-D LINE ELEMENT	36	SIZE	0.250	LENGTH	5.500	AREA	0.972
56	STRESS AT NODES 143,168			5.4,	8.4	ALLOWABLE		27.3
57	STRESS EXPANSION FOR NODE 143							
	1.1	0.9		-8.0	0.9		0.0	0.0
	STRESS EXPANSION FOR NODE 168							
	1.5	0.9		-11.2	0.9		0.0	0.0

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LAEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* SSE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

3-D LINE ELEMENT	37	SIZE	0.250	LENGTH	6.000	AREA	1.061
STRESS AT NODES 161,162			12.9		9.5	ALLOWABLE	27.3
STRESS EXPANSION FOR NODE 161							
	1.6	-0.9	-11.5		-3.8	0.0	0.0
STRESS EXPANSION FOR NODE 162							
	1.1	-0.9	-8.0		-3.8	0.0	0.0

3-D LINE ELEMENT	38	SIZE	0.250	LENGTH	6.000	AREA	1.061
STRESS AT NODES 163,164			13.2		9.8	ALLOWABLE	27.3
STRESS EXPANSION FOR NODE 163							
	1.6	0.9	-11.5		-3.8	0.0	0.0
STRESS EXPANSION FOR NODE 164							
	1.1	0.9	-8.0		-3.8	0.0	0.0

FORCE DEFINITION	NODE	DIRECT	ELEMENT	39	NODE	200	
FX	22.20	;	FY	78.80	;	FZ	0.00
MX	-5384.00	;	MY	0.00	;	MZ	0.00

MAXIMUM ABSOLUTE STRESS ON ELEMENT 38 = 13.2 KSI

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AEP \* 150T CRANE \* 50T LOAD \* TRUCK TO GIRDER \* WELDS \* SSE

\*\*\*\*\* REPORT SUMMARY, FOR A 4 LOAD STEP ANALYSIS \*\*\*\*\*

WORST CASE ANALYSIS ..... NO

ELEMENT STRESSES PRINTED... YES, STRESS SUMMATION MODE WAS DIRECT

ELEMENT STATISTICS

TYPE	DESCRIPTION	NUMBER USED
1	3-DIMENSIONAL LINE ELEMENT.....	36
2	3-DIMENSIONAL LINE ELEMENT.....	2
3	FORCE DEFINITION NODE - DIRECT.....	1

TOTAL NUMBER OF ELEMENTS = 39, NODES = 69

SYSTEM PROPERTIES

LOAD STEP..... 1 DIRECTION, LOCATION OR AXIS

	X	Y	Z
CENTROIDS, X AXIS.....		0.00	22.50
Y AXIS.....	0.00		22.50
Z AXIS.....	-19.99	-0.00	
SHEAR AREAS.....	33.23	33.23	0.00
POLAR MOMENTS OF INERTIA..	6299.12	6299.12	15729.29
TRANSLATED FORCES.....	22.20	78.80	0.00
TRANSLATED MOMENTS.....	-3611.08	-499.48	1574.85

MAXIMUM ELEMENT STRESSES

DESCRIPTION	NUMBER USED	ELEMENT	NODE	STRESS	ALLOWABLE	COMPARISON FACTOR
3-D LINE ELEMENT	36	38	163	13.2 KSI	27.3 KSI	2.064
3-D LINE ELEMENT	2	1	101	12.8 KSI	27.3 KSI	2.138
3-D LINE ELEMENT	36	38	163	13.2 KSI	27.3 KSI	2.064
3-D LINE ELEMENT	2	1	101	12.8 KSI	27.3 KSI	2.138

\*\*\*\*\* JOB COMPLETED \*\*\*\*\*

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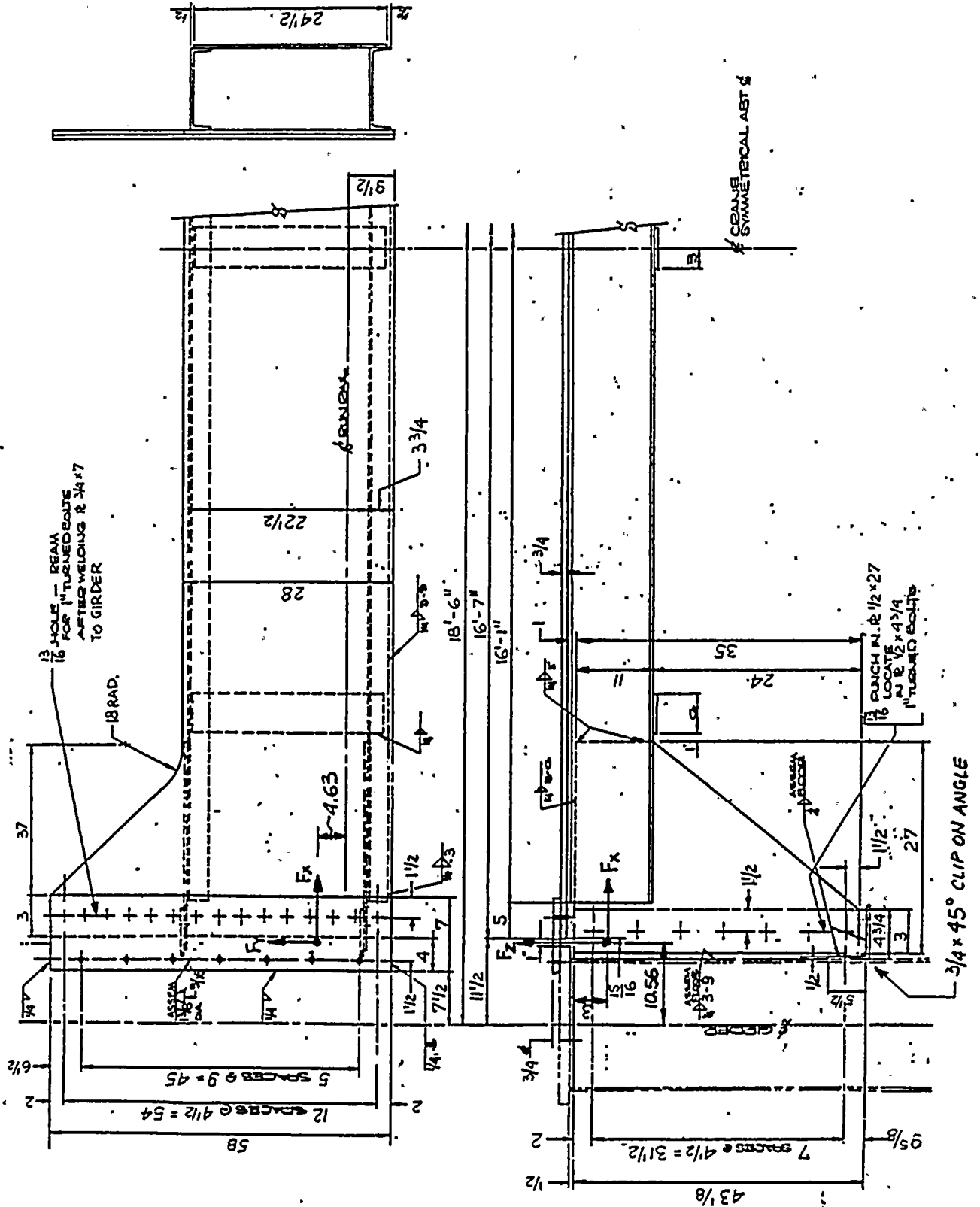
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WHITING REQN. 79604 DATE 6-26-87  
 BY ASZ PAGE 4-71 OF 112  
 MJM 8-26-87

GIRDER TO END TIE CONNECTION.





WHITING REQN. 79604 DATE 6-26-37  
 BY ASZ PAGE 4-72 OF 42  
 MJM 8-26-87

ALLOWABLES

1" BOLTS

MTRL: ASTM-A325

$$\begin{array}{ll} \text{OBE } \sigma_{ALL} = 61.3 \text{ KSI} & \tau_{ALL} = 36.8 \text{ KSI} \\ \text{SSE } \sigma_{ALL} = 83.6 \text{ KSI} & \tau_{ALL} = 50.2 \text{ KSI} \end{array} \rightarrow (\text{REF. PG. 4-38})$$

WELDS (FILLET THRU THROAT)

BASE MTRL: ASTM-A36

WELD MTRL: MIN. E60XX ELECTRODES

$$\begin{array}{ll} \text{OBE } \tau_{W,ALL} = 20.0 \text{ KSI} & \\ \text{SSE } \tau_{W,ALL} = 27.3 \text{ KSI} & \rightarrow (\text{REF. PG. 4-38}) \end{array}$$

MAX. LOADS PER TABLE B55 AND B56.

$$\text{OBE } F_x = 38.5 \text{ KIP } F_y = 34 \text{ KIP } F_z = 2.1 \text{ KIP } M_x = 8 \text{ IN. KIP. } M_y = 87.1 \text{ IN. KIP.}$$

$$M_z = 3731 \text{ IN. KIP}$$

$$\text{SSE } F_x = 57.7 \text{ KIP } F_y = 49.5 \text{ KIP } F_z = 2.2 \text{ KIP } M_x = 10.5 \text{ IN. KIP}$$

$$M_y = 103.3 \text{ IN. KIP } M_z = 6359 \text{ IN. KIP.}$$

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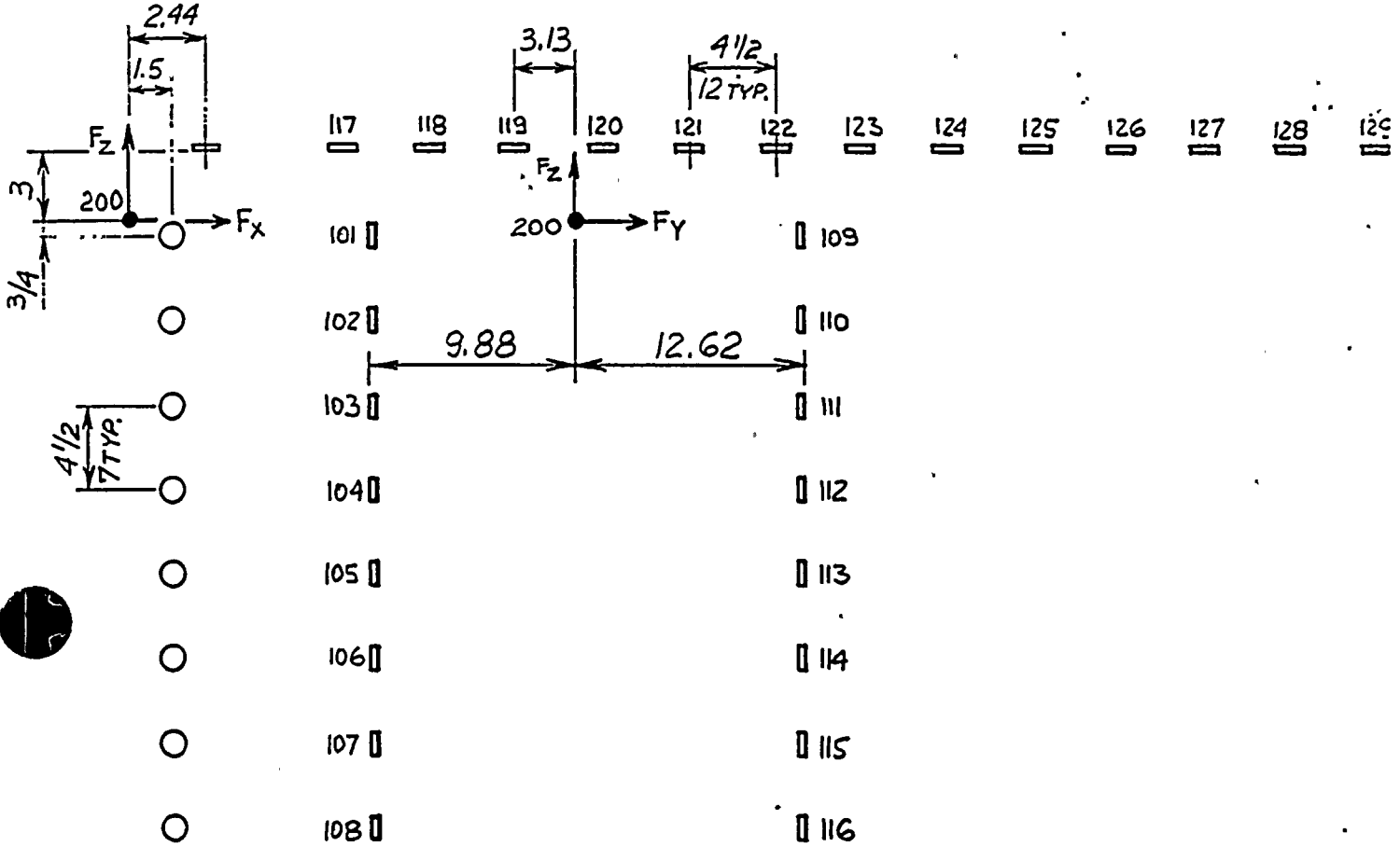
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WHITING REQ. 79604 DATE 6-26-87  
 BY ASZ PAGE 4-73 OF 112  
 MJM 8-26-87

BOLTED CONNECTION



ALL 7" A325 BOLTS

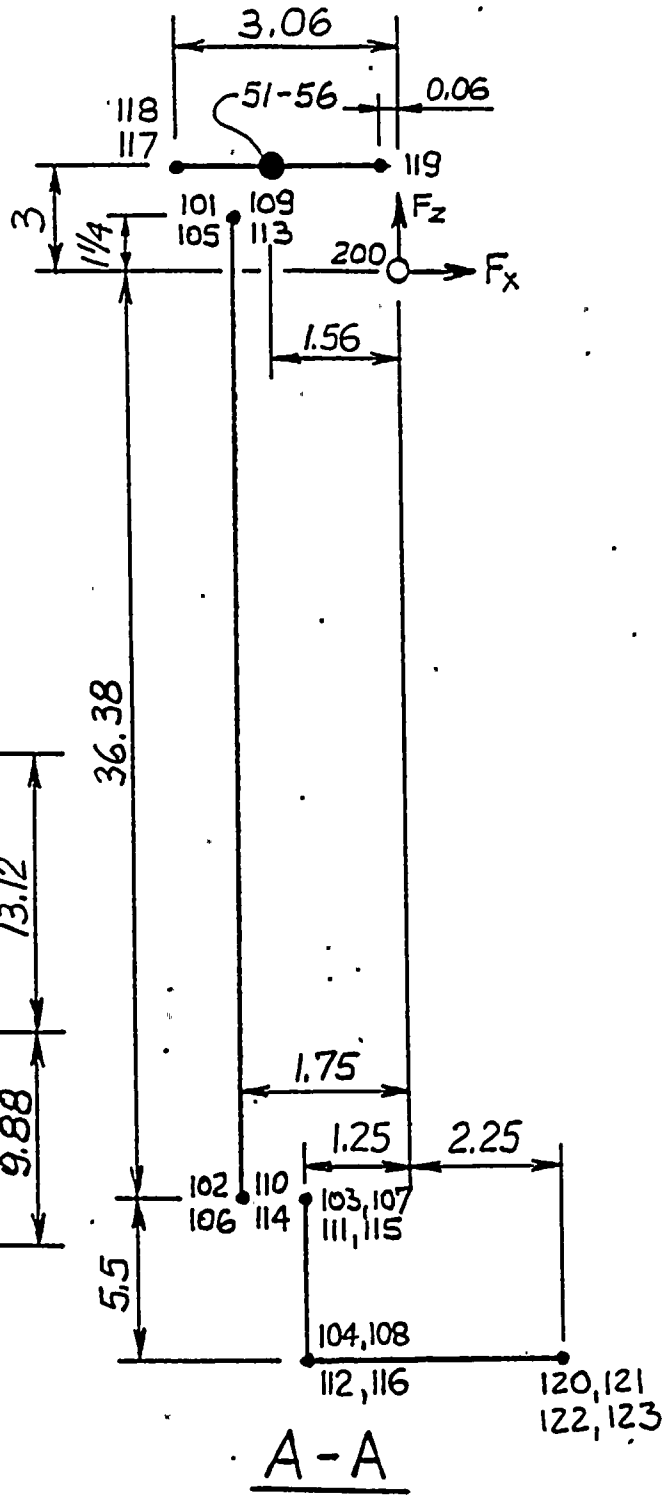
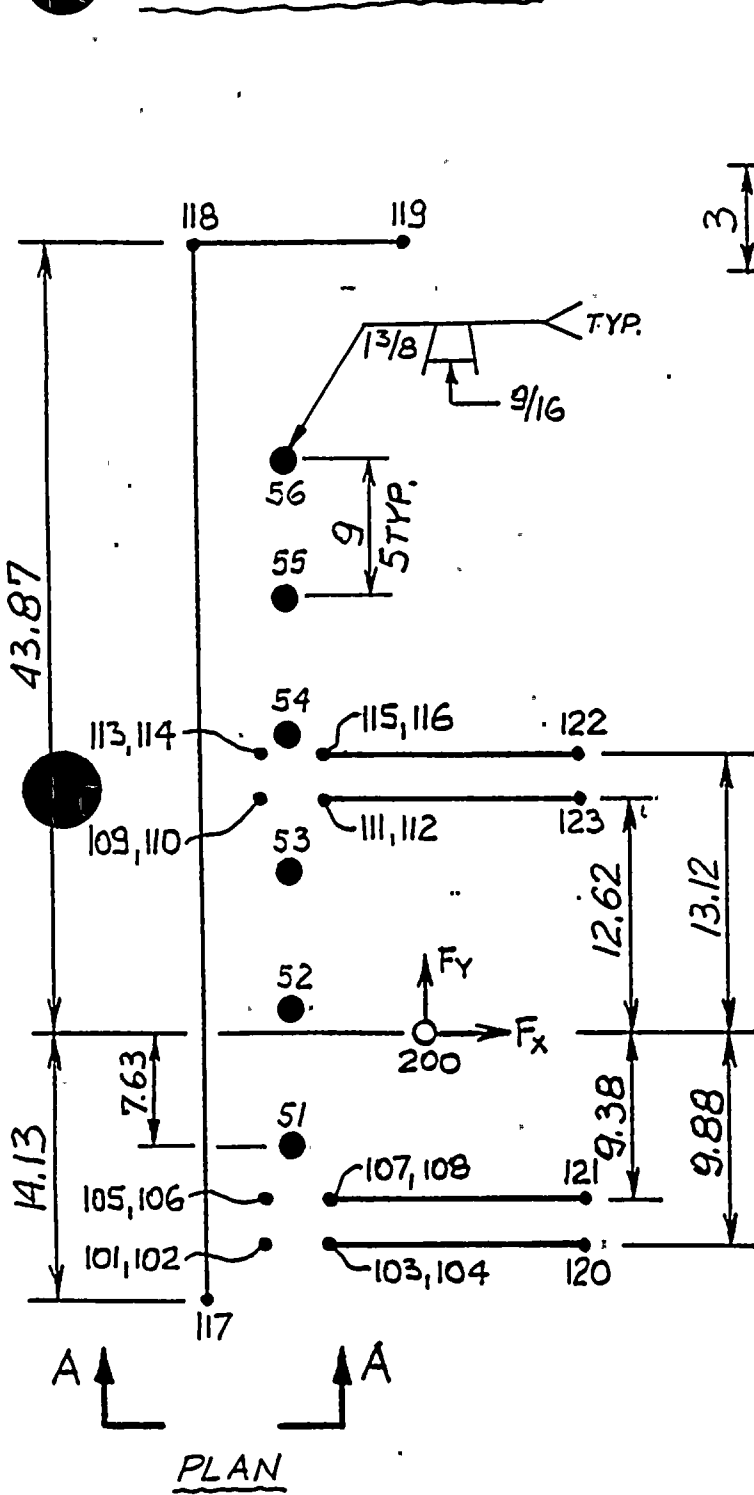
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WELDED CONNECTION.



1/4 FILLET WELDS EXCEPT AS NOTED

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AEP \* 150T CRANE \* 50T LOAD \* GIRDER TO END TIE \* BOLTS \* OBE

\*\*\*\*\* SYSTEM PROPERTIES \*\*\*\*\*

LOAD STEP.....	1	DIRECTION, LOCATION OR AXIS		
		X	Y	Z
CENTROIDS, X AXIS.....			1.37	3.00
Y AXIS.....		1.50		-7.76
Z AXIS.....		2.44	7.42	
SHEAR AREAS.....		22.78	10.21	12.57
POLAR MOMENTS OF INERTIA..		1590.43	3477.98	5511.66
TRANSLATED FORCES.....		38.50	34.00	2.10
TRANSLATED MOMENTS.....		112.88	388.96	4099.70
NUMBER OF FORCE DEFINITION NODES...	1			

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

DIRECT STRESSES ARE EQUALLY DISTRIBUTED THROUGHOUT SHEAR AREA  
 DRX 1.7 DRY 3.3 DRZ 0.2

2-D POINT ELEMENT 1	NODE 101	AREA 0.785	DIAMETER 1.000
STRESS AT NODE 15.4	ALLOWABLE 36.8		
STRESS EXPANSION FOR NODE 101			
0.8	12.9	0.0	0.0
FORCES AT NODE 101	FX 12.	FY 0.	FZ -0.8
			1. 0.0
2-D POINT ELEMENT 2	NODE 102	AREA 0.785	DIAMETER 1.000
STRESS AT NODE 14.9	ALLOWABLE 36.8		
STRESS EXPANSION FOR NODE 102			
0.3	12.9	0.0	0.0
FORCES AT NODE 102	FX 12.	FY 0.	FZ -0.8
			1. 0.0
2-D POINT ELEMENT 3	NODE 103	AREA 0.785	DIAMETER 1.000
STRESS AT NODE 14.8	ALLOWABLE 36.8		
STRESS EXPANSION FOR NODE 103			
-0.2	12.9	0.0	0.0
FORCES AT NODE 103	FX 12.	FY 0.	FZ -0.8
			1. 0.0
2-D POINT ELEMENT 4	NODE 104	AREA 0.785	DIAMETER 1.000
STRESS AT NODE 15.3	ALLOWABLE 36.8		
STRESS EXPANSION FOR NODE 104			
-0.7	12.9	0.0	0.0
FORCES AT NODE 104	FX 12.	FY 0.	FZ -0.8
			1. 0.0
2-D POINT ELEMENT 5	NODE 105	AREA 0.785	DIAMETER 1.000
STRESS AT NODE 15.8	ALLOWABLE 36.8		
STRESS EXPANSION FOR NODE 105			
-1.2	12.9	0.0	0.0
FORCES AT NODE 105	FX 12.	FY 0.	FZ -0.8
			1. 0.0

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176

176



AEP \* 150T CRANE \* 50T LOAD \* GIRDER TO END TIE \* BOLTS \* OBE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

6	2-D POINT ELEMENT	6	NODE 106	AREA	0.785	DIAMETER	1.000
7	STRESS AT NODE	16.3	ALLOWABLE	36.8			
8	STRESS EXPANSION FOR NODE 106						
9		-1.7	12.9	0.0	0.0	-0.8	0.0
10	FORCES AT NODE 106	FX	13.	FY	0.	FZ	1.
12	2-D POINT ELEMENT	7	NODE 107	AREA	0.785	DIAMETER	1.000
13	STRESS AT NODE	16.8	ALLOWABLE	36.8			
14	STRESS EXPANSION FOR NODE 107						
15		-2.2	12.9	0.0	0.0	-0.8	0.0
16	FORCES AT NODE 107	FX	13.	FY	0.	FZ	1.
18	2-D POINT ELEMENT	8	NODE 108	AREA	0.785	DIAMETER	1.000
19	STRESS AT NODE	17.3	ALLOWABLE	36.8			
20	STRESS EXPANSION FOR NODE 108						
21		-2.7	12.9	0.0	0.0	-0.8	0.0
22	FORCES AT NODE 108	FX	14.	FY	0.	FZ	1.
24	2-D POINT ELEMENT	9	NODE 109	AREA	0.785	DIAMETER	1.000
25	STRESS AT NODE	6.4	ALLOWABLE	36.8			
26	STRESS EXPANSION FOR NODE 109						
27		0.8	-3.9	0.0	0.0	0.8	0.0
28	FORCES AT NODE 109	FX	5.	FY	0.	FZ	1.
30	2-D POINT ELEMENT	10	NODE 110	AREA	0.785	DIAMETER	1.000
31	STRESS AT NODE	5.9	ALLOWABLE	36.8			
32	STRESS EXPANSION FOR NODE 110						
33		0.3	-3.9	0.0	0.0	0.8	0.0
34	FORCES AT NODE 110	FX	5.	FY	0.	FZ	1.
36	2-D POINT ELEMENT	11	NODE 111	AREA	0.785	DIAMETER	1.000
37	STRESS AT NODE	5.9	ALLOWABLE	36.8			
38	STRESS EXPANSION FOR NODE 111						
39		-0.2	-3.9	0.0	0.0	0.8	0.0
40	FORCES AT NODE 111	FX	5.	FY	0.	FZ	1.
42	2-D POINT ELEMENT	12	NODE 112	AREA	0.785	DIAMETER	1.000
43	STRESS AT NODE	6.4	ALLOWABLE	36.8			
44	STRESS EXPANSION FOR NODE 112						
45		-0.7	-3.9	0.0	0.0	0.8	0.0
46	FORCES AT NODE 112	FX	5.	FY	0.	FZ	1.
48	2-D POINT ELEMENT	13	NODE 113	AREA	0.785	DIAMETER	1.000
49	STRESS AT NODE	6.9	ALLOWABLE	36.8			
50	STRESS EXPANSION FOR NODE 113						
51		-1.2	-3.9	0.0	0.0	0.8	0.0
52	FORCES AT NODE 113	FX	5.	FY	0.	FZ	1.
54	2-D POINT ELEMENT	14	NODE 114	AREA	0.785	DIAMETER	1.000
55	STRESS AT NODE	7.4	ALLOWABLE	36.8			
56	STRESS EXPANSION FOR NODE 114						
57		-1.7	-3.9	0.0	0.0	0.8	0.0
	FORCES AT NODE 114	FX	6.	FY	0.	FZ	1.

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AEP \* 150T CRANE \* 50T LOAD \* GIRDER TO END TIE \* BOLTS \* OBE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

2-D POINT ELEMENT 15 NODE 115 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 7.9 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 115  
 -2.2 -3.9 0.0 0.0 0.8 0.0  
 FORCES AT NODE 115 FX 6. FY 0. FZ 1.

2-D POINT ELEMENT 16 NODE 116 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 8.4 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 116  
 -2.7 -3.9 0.0 0.0 0.8 0.0  
 FORCES AT NODE 116 FX 7. FY 0. FZ 1.

2-D POINT ELEMENT 17 NODE 117 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 17.8 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 117  
 1.2 14.5 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 117 FX 14. FY 3. FZ 0.

2-D POINT ELEMENT 18 NODE 118 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 14.5 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 118  
 1.2 11.2 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 118 FX 11. FY 3. FZ 0.

2-D POINT ELEMENT 19 NODE 119 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 11.2 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 119  
 1.2 7.8 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 119 FX 8. FY 3. FZ 0.

2-D POINT ELEMENT 20 NODE 120 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 8.1 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 120  
 1.2 4.5 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 120 FX 6. FY 3. FZ 0.

2-D POINT ELEMENT 21 NODE 121 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 5.2 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 121  
 1.2 1.2 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 121 FX 3. FY 3. FZ 0.

2-D POINT ELEMENT 22 NODE 122 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 6.1 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 122  
 1.2 -2.2 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 122 FX 4. FY 3. FZ 0.

2-D POINT ELEMENT 23 NODE 123 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 9.1 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 123  
 1.2 -5.5 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 123 FX 7. FY 3. FZ 0.

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AEP \* 150T CRANE \* 50T LOAD \* GIRDER TO END TIE \* BOLTS \* OBE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

2-D POINT ELEMENT 24 NODE 124 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 12.2 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 124  
 1.2 -8.9 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 124 FX 9. FY 3. FZ 0.

2-D POINT ELEMENT 25 NODE 125 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 15.5 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 125  
 1.2 -12.2 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 125 FX 12. FY 3. FZ 0.

2-D POINT ELEMENT 26 NODE 126 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 18.8 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 126  
 1.2 -15.6 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 126 FX 15. FY 3. FZ 0.

2-D POINT ELEMENT 27 NODE 127 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 22.1 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 127  
 1.2 -18.9 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 127 FX 17. FY 3. FZ 0.

2-D POINT ELEMENT 28 NODE 128 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 25.4 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 128  
 1.2 -22.3 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 128 FX 20. FY 3. FZ 0.

2-D POINT ELEMENT 29 NODE 129 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 28.7 ALLOWABLE 36.8  
 STRESS EXPANSION FOR NODE 129  
 1.2 -25.6 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 129 FX 22. FY 3. FZ 0.

FORCE DEFINITION NODE ABSOLUTE ELEMENT 30 NODE 200  
 FX 38.50 ; FY 34.00 ; FZ 2.10  
 MX 8.00 ; MY 87.10 ; MZ 3731.00

MAXIMUM ABSOLUTE STRESS ON ELEMENT 29 = 28.7 KSI

\*\*\*\*\* JOB COMPLETED \*\*\*\*\*

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\*\*\*\*\* SYSTEM PROPERTIES \*\*\*\*\*

LOAD STEP..... 1	DIRECTION, LOCATION OR AXIS		
	X	Y	Z
CENTROIDS, X AXIS.....		1.37	3.00
Y AXIS.....	1.50		-7.76
Z AXIS.....	2.44	7.42	
SHEAR AREAS.....	22.78	10.21	12.57
POLAR MOMENTS OF INERTIA..	1590.43	3477.98	5511.66
TRANSLATED FORCES.....	57.70	49.50	2.20
TRANSLATED MOMENTS.....	162.01	554.27	6908.01
NUMBER OF FORCE DEFINITION NODES... 1			

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

DIRECT STRESSES ARE EQUALLY DISTRIBUTED THROUGHOUT SHEAR AREA  
 DRX 2.5 DRY 4.8 DRZ 0.2

2-D POINT ELEMENT 1	NODE 101	AREA 0.785	DIAMETER 1.000
STRESS AT NODE	25.4	ALLOWABLE 50.2	
STRESS EXPANSION FOR NODE 101			
1.1	21.7	0.0	0.0
FORCES AT NODE 101	FX 20.	FY 0.	FZ -1.1
			1. 0.0
2-D POINT ELEMENT 2	NODE 102	AREA 0.785	DIAMETER 1.000
STRESS AT NODE	24.7	ALLOWABLE 50.2	
STRESS EXPANSION FOR NODE 102			
0.4	21.7	0.0	0.0
FORCES AT NODE 102	FX 19.	FY 0.	FZ -1.1
			1. 0.0
2-D POINT ELEMENT 3	NODE 103	AREA 0.785	DIAMETER 1.000
STRESS AT NODE	24.6	ALLOWABLE 50.2	
STRESS EXPANSION FOR NODE 103			
-0.3	21.7	0.0	0.0
FORCES AT NODE 103	FX 19.	FY 0.	FZ -1.1
			1. 0.0
2-D POINT ELEMENT 4	NODE 104	AREA 0.785	DIAMETER 1.000
STRESS AT NODE	25.3	ALLOWABLE 50.2	
STRESS EXPANSION FOR NODE 104			
-1.0	21.7	0.0	0.0
FORCES AT NODE 104	FX 20.	FY 0.	FZ -1.1
			1. 0.0
2-D POINT ELEMENT 5	NODE 105	AREA 0.785	DIAMETER 1.000
STRESS AT NODE	26.0	ALLOWABLE 50.2	
STRESS EXPANSION FOR NODE 105			
-1.8	21.7	0.0	0.0
FORCES AT NODE 105	FX 20.	FY 0.	FZ -1.1
			1. 0.0

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\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

2-D POINT ELEMENT 6 NODE 106 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 26.7 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 106  
 -2.5 21.7 0.0 0.0 -1.1 0.0  
 FORCES AT NODE 106 FX 21. FY 0. FZ 1.

2-D POINT ELEMENT 7 NODE 107 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 27.4 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 107  
 -3.2 21.7 0.0 0.0 -1.1 0.0  
 FORCES AT NODE 107 FX 22. FY 0. FZ 1.

2-D POINT ELEMENT 8 NODE 108 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 28.2 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 108  
 -3.9 21.7 0.0 0.0 -1.1 0.0  
 FORCES AT NODE 108 FX 22. FY 0. FZ 1.

2-D POINT ELEMENT 9 NODE 109 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 10.3 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 109  
 1.1 -6.5 0.0 0.0 1.1 0.0  
 FORCES AT NODE 109 FX 8. FY 0. FZ 1.

2-D POINT ELEMENT 10 NODE 110 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 9.5 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 110  
 0.4 -6.5 0.0 0.0 1.1 0.0  
 FORCES AT NODE 110 FX 7. FY 0. FZ 1.

2-D POINT ELEMENT 11 NODE 111 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 9.5 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 111  
 -0.3 -6.5 0.0 0.0 1.1 0.0  
 FORCES AT NODE 111 FX 7. FY 0. FZ 1.

2-D POINT ELEMENT 12 NODE 112 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 10.2 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 112  
 -1.0 -6.5 0.0 0.0 1.1 0.0  
 FORCES AT NODE 112 FX 8. FY 0. FZ 1.

2-D POINT ELEMENT 13 NODE 113 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 10.9 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 113  
 -1.8 -6.5 0.0 0.0 1.1 0.0  
 FORCES AT NODE 113 FX 8. FY 0. FZ 1.

2-D POINT ELEMENT 14 NODE 114 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 11.6 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 114  
 -2.5 -6.5 0.0 0.0 1.1 0.0  
 FORCES AT NODE 114 FX 9. FY 0. FZ 1.

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\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

6	2-D POINT ELEMENT	15	NODE 115	AREA	0.785	DIAMETER	1.000
7	STRESS AT NODE	12.3	ALLOWABLE	50.2			
8	STRESS EXPANSION FOR NODE 115						
9		-3.2	-6.5	0.0	0.0	1.1	0.0
10	FORCES AT NODE 115	FX	10.	FY	0.	FZ	1.
12	2-D POINT ELEMENT	16	NODE 116	AREA	0.785	DIAMETER	1.000
13	STRESS AT NODE	13.0	ALLOWABLE	50.2			
14	STRESS EXPANSION FOR NODE 116						
15		-3.9	-6.5	0.0	0.0	1.1	0.0
16	FORCES AT NODE 116	FX	10.	FY	0.	FZ	1.
18	2-D POINT ELEMENT	17	NODE 117	AREA	0.785	DIAMETER	1.000
19	STRESS AT NODE	29.2	ALLOWABLE	50.2			
20	STRESS EXPANSION FOR NODE 117						
21		1.7	24.5	0.0	-0.0	0.0	0.0
22	FORCES AT NODE 117	FX	23.	FY	4.	FZ	0.
24	2-D POINT ELEMENT	18	NODE 118	AREA	0.785	DIAMETER	1.000
25	STRESS AT NODE	23.6	ALLOWABLE	50.2			
26	STRESS EXPANSION FOR NODE 118						
27		1.7	18.9	0.0	-0.0	0.0	0.0
28	FORCES AT NODE 118	FX	18.	FY	4.	FZ	0.
30	2-D POINT ELEMENT	19	NODE 119	AREA	0.785	DIAMETER	1.000
31	STRESS AT NODE	18.1	ALLOWABLE	50.2			
32	STRESS EXPANSION FOR NODE 119						
33		1.7	13.2	0.0	-0.0	0.0	0.0
34	FORCES AT NODE 119	FX	14.	FY	4.	FZ	0.
36	2-D POINT ELEMENT	20	NODE 120	AREA	0.785	DIAMETER	1.000
37	STRESS AT NODE	12.8	ALLOWABLE	50.2			
38	STRESS EXPANSION FOR NODE 120						
39		1.7	7.6	0.0	-0.0	0.0	0.0
40	FORCES AT NODE 120	FX	9.	FY	4.	FZ	0.
42	2-D POINT ELEMENT	21	NODE 121	AREA	0.785	DIAMETER	1.000
43	STRESS AT NODE	7.9	ALLOWABLE	50.2			
44	STRESS EXPANSION FOR NODE 121						
45		1.7	1.9	0.0	-0.0	0.0	0.0
46	FORCES AT NODE 121	FX	5.	FY	4.	FZ	0.
48	2-D POINT ELEMENT	22	NODE 122	AREA	0.785	DIAMETER	1.000
49	STRESS AT NODE	9.3	ALLOWABLE	50.2			
50	STRESS EXPANSION FOR NODE 122						
51		1.7	-3.7	0.0	-0.0	0.0	0.0
52	FORCES AT NODE 122	FX	6.	FY	4.	FZ	0.
54	2-D POINT ELEMENT	23	NODE 123	AREA	0.785	DIAMETER	1.000
55	STRESS AT NODE	14.4	ALLOWABLE	50.2			
56	STRESS EXPANSION FOR NODE 123						
57		1.7	-9.3	0.0	-0.0	0.0	0.0
	FORCES AT NODE 123	FX	11.	FY	4.	FZ	0.

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\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

2-D POINT ELEMENT 24 NODE 124 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 19.8 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 124  
 1.7 -15.0 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 124 FX 15. FY 4. FZ 0.

2-D POINT ELEMENT 25 NODE 125 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 25.3 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 125  
 1.7 -20.6 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 125 FX 20. FY 4. FZ 0.

2-D POINT ELEMENT 26 NODE 126 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 30.9 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 126  
 1.7 -26.3 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 126 FX 24. FY 4. FZ 0.

2-D POINT ELEMENT 27 NODE 127 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 36.5 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 127  
 1.7 -31.9 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 127 FX 28. FY 4. FZ 0.

2-D POINT ELEMENT 28 NODE 128 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 42.1 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 128  
 1.7 -37.5 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 128 FX 33. FY 4. FZ 0.

2-D POINT ELEMENT 29 NODE 129 AREA 0.785 DIAMETER 1.000  
 STRESS AT NODE 47.7 ALLOWABLE 50.2  
 STRESS EXPANSION FOR NODE 129  
 1.7 -43.2 0.0 -0.0 0.0 0.0  
 FORCES AT NODE 129 FX 37. FY 4. FZ 0.

FORCE DEFINITION NODE ABSOLUTE ELEMENT 30 NODE 200  
 FX 57.70 ; FY 49.50 ; FZ 2.20  
 MX 10.50 ; MY 103.30 ; MZ 6359.00

MAXIMUM ABSOLUTE STRESS ON ELEMENT 29 = 47.7 KSI

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PAEP \* 150T CRANE \* 50T LOAD \* GIRDER TO END TIE \* WELDS \* OBE

\*\*\*\*\* SYSTEM PROPERTIES \*\*\*\*\*

LOAD STEP..... 1	DIRECTION, LOCATION OR AXIS		
	X	Y	Z
CENTROIDS, X AXIS.....		5.24	-12.61
Y AXIS.....	-1.88		-12.61
Z AXIS.....	-1.83	6.87	
SHEAR AREAS.....	52.67	52.67	43.76
POLAR MOMENTS OF INERTIA..	22680.72	13490.29	12049.47
TRANSLATED FORCES.....	38.50	34.00	2.10
TRANSLATED MOMENTS.....	447.75	576.55	4057.52
NUMBER OF FORCE DEFINITION NODES... 1			

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

DIRECT STRESSES ARE EQUALLY DISTRIBUTED THROUGHOUT SHEAR AREA  
 DRX 0.7 DRZ 0.6 DRZ 0.0

3-D LINE ELEMENT 1	SIZE 0.250	LENGTH 37.630	AREA 6.652
STRESS AT NODES 101,102	7.0,	7.5	ALLOWABLE 20.0
STRESS EXPANSION FOR NODE 101	0.6	5.6	-0.3 0.0 -0.3 -0.0
STRESS EXPANSION FOR NODE 102	-1.0	5.6	0.5 0.0 -0.3 -0.0
3-D LINE ELEMENT 2	SIZE 0.250	LENGTH 5.500	AREA 0.972
STRESS AT NODES 103,104	7.5,	7.8	ALLOWABLE 20.0
STRESS EXPANSION FOR NODE 103	-1.0	5.6	0.5 0.2 -0.3 -0.0
STRESS EXPANSION FOR NODE 104	-1.3	5.6	0.6 0.2 -0.3 -0.0
3-D LINE ELEMENT 3	SIZE 0.250	LENGTH 37.630	AREA 6.652
STRESS AT NODES 105,106	6.9,	7.3	ALLOWABLE 20.0
STRESS EXPANSION FOR NODE 105	0.6	5.5	-0.3 0.0 -0.3 -0.0
STRESS EXPANSION FOR NODE 106	-1.0	5.5	0.5 0.0 -0.3 -0.0
3-D LINE ELEMENT 4	SIZE 0.250	LENGTH 5.500	AREA 0.972
STRESS AT NODES 107,108	7.3,	7.6	ALLOWABLE 20.0
STRESS EXPANSION FOR NODE 107	-1.0	5.5	0.5 0.2 -0.3 -0.0
STRESS EXPANSION FOR NODE 108	-1.3	5.5	0.6 0.2 -0.3 -0.0

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MJM 8-22-87

AEP \* 150T CRANE \* 50T LOAD \* GIRDER TO END TIE \* WELDS \* OBE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

3-D LINE ELEMENT	5	SIZE	0.250	LENGTH	37.630	AREA	6.652
STRESS AT NODES 109,110			3.4,	3.9	ALLOWABLE		20.0
STRESS EXPANSION FOR NODE 109	0.6	-1.9	-0.3	0.0	0.1		-0.0
STRESS EXPANSION FOR NODE 110	-1.0	-1.9	0.5	0.0	0.1		-0.0
3-D LINE ELEMENT	6	SIZE	0.250	LENGTH	5.500	AREA	0.972
STRESS AT NODES 111,112			3.9,	4.2	ALLOWABLE		20.0
STRESS EXPANSION FOR NODE 111	-1.0	-1.9	0.5	0.2	0.1		-0.0
STRESS EXPANSION FOR NODE 112	-1.3	-1.9	0.6	0.2	0.1		-0.0
3-D LINE ELEMENT	7	SIZE	0.250	LENGTH	37.630	AREA	6.652
STRESS AT NODES 113,114			3.6,	4.0	ALLOWABLE		20.0
STRESS EXPANSION FOR NODE 113	0.6	-2.1	-0.3	0.0	0.2		-0.0
STRESS EXPANSION FOR NODE 114	-1.0	-2.1	0.5	0.0	0.2		-0.0
3-D LINE ELEMENT	8	SIZE	0.250	LENGTH	5.500	AREA	0.972
STRESS AT NODES 115,116			4.1,	4.3	ALLOWABLE		20.0
STRESS EXPANSION FOR NODE 115	-1.0	-2.1	0.5	0.2	0.2		-0.0
STRESS EXPANSION FOR NODE 116	-1.3	-2.1	0.6	0.2	0.2		-0.0
3-D LINE ELEMENT	9	SIZE	0.250	LENGTH	58.000	AREA	10.253
STRESS AT NODES 117,118			8.6,	14.0	ALLOWABLE		20.0
STRESS EXPANSION FOR NODE 117	0.7	7.1	-0.3	-0.4	-0.4		0.1
STRESS EXPANSION FOR NODE 118	0.7	-12.5	-0.3	-0.4	0.8		0.1
3-D LINE ELEMENT	10	SIZE	0.250	LENGTH	3.000	AREA	0.530
STRESS AT NODES 118,119			14.0,	14.0	ALLOWABLE		20.0
STRESS EXPANSION FOR NODE 118	0.7	-12.5	-0.3	-0.4	0.8		0.1
STRESS EXPANSION FOR NODE 119	0.7	-12.5	-0.3	0.6	0.8		-0.1
3-D LINE ELEMENT	11	SIZE	0.250	LENGTH	3.500	AREA	0.619
STRESS AT NODES 104,120			7.8,	8.1	ALLOWABLE		20.0
STRESS EXPANSION FOR NODE 104	-1.3	5.6	0.6	0.2	-0.3		-0.0
STRESS EXPANSION FOR NODE 120	-1.3	5.6	0.6	1.4	-0.3		-0.2
3-D LINE ELEMENT	12	SIZE	0.250	LENGTH	3.500	AREA	0.619
STRESS AT NODES 108,121			7.6,	7.9	ALLOWABLE		20.0
STRESS EXPANSION FOR NODE 108	-1.3	5.5	0.6	0.2	-0.3		-0.0
STRESS EXPANSION FOR NODE 121	-1.3	5.5	0.6	1.4	-0.3		-0.2

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AEP \* 150T CRANE \* 50T LOAD \* GIRDER TO END TIE \* WELDS \* OBE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

3-D LINE ELEMENT 13 SIZE 0.250 LENGTH 3.500 AREA 0.619  
 STRESS AT NODES 116,122 4.3, 4.9 ALLOWABLE 20.0  
 STRESS EXPANSION FOR NODE 116  
 -1.3 -2.1 0.6 0.2 0.2 -0.0  
 STRESS EXPANSION FOR NODE 122  
 -1.3 -2.1 0.6 1.4 0.2 -0.2

3-D LINE ELEMENT 14 SIZE 0.250 LENGTH 3.500 AREA 0.619  
 STRESS AT NODES 112,123 4.2, 4.7 ALLOWABLE 20.0  
 STRESS EXPANSION FOR NODE 112  
 -1.3 -1.9 0.6 0.2 0.1 -0.0  
 STRESS EXPANSION FOR NODE 123  
 -1.3 -1.9 0.6 1.4 0.1 -0.2

2-D POINT ELEMENT 15 NODE 51 AREA 1.485 DIAMETER 1.375  
 STRESS AT NODE 6.4 ALLOWABLE 20.0  
 STRESS EXPANSION FOR NODE 51  
 0.7 4.9 -0.3 0.1 0.0 0.0

2-D POINT ELEMENT 16 NODE 52 AREA 1.485 DIAMETER 1.375  
 STRESS AT NODE 3.4 ALLOWABLE 20.0  
 STRESS EXPANSION FOR NODE 52  
 0.7 1.9 -0.3 0.1 0.0 0.0

2-D POINT ELEMENT 17 NODE 53 AREA 1.485 DIAMETER 1.375  
 STRESS AT NODE 2.8 ALLOWABLE 20.0  
 STRESS EXPANSION FOR NODE 53  
 0.7 -1.2 -0.3 0.1 0.0 0.0

2-D POINT ELEMENT 18 NODE 54 AREA 1.485 DIAMETER 1.375  
 STRESS AT NODE 5.7 ALLOWABLE 20.0  
 STRESS EXPANSION FOR NODE 54  
 0.7 -4.2 -0.3 0.1 0.0 0.0

2-D POINT ELEMENT 19 NODE 55 AREA 1.485 DIAMETER 1.375  
 STRESS AT NODE 8.7 ALLOWABLE 20.0  
 STRESS EXPANSION FOR NODE 55  
 0.7 -7.2 -0.3 0.1 0.0 0.0

2-D POINT ELEMENT 20 NODE 56 AREA 1.485 DIAMETER 1.375  
 STRESS AT NODE 11.7 ALLOWABLE 20.0  
 STRESS EXPANSION FOR NODE 56  
 0.7 -10.3 -0.3 0.1 0.0 0.0

FORCE DEFINITION NODE ABSOLUTE ELEMENT 21 NODE 200  
 FX 38.50 ; FY 34.00 ; FZ 2.10  
 MX 8.00 ; NY 87.10 ; MZ 3731.00

MAXIMUM ABSOLUTE STRESS ON ELEMENT 10 = 14.0 KSI

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AEP \* 150T CRANE \* 50T LOAD \* GIRDER TO END TIE \* WELDS \* SSE

\*\*\*\*\* SYSTEM PROPERTIES \*\*\*\*\*

LOAD STEP	DIRECTION, LOCATION OR AXIS		
	X	Y	Z
CENTROIDS, X AXIS.....		5.24	-12.61
Y AXIS.....	-1.88		-12.61
Z AXIS.....	-1.83	6.87	
SHEAR AREAS.....	52.67	52.67	43.76
POLAR MOMENTS OF INERTIA..	22680.72	13490.29	12049.47
TRANSLATED FORCES.....	57.70	49.50	2.20
TRANSLATED MOMENTS.....	646.24	835.06	6845.70
NUMBER OF FORCE DEFINITION NODES... 1			

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

DIRECT STRESSES ARE EQUALLY DISTRIBUTED THROUGHOUT SHEAR AREA  
 DRX 1.1 DRY 0.9 DRZ 0.1

3-D LINE ELEMENT 1	SIZE 0.250	LENGTH 37.630	AREA 6.652
STRESS AT NODES 101, 102	11.6,	12.2	ALLOWABLE 27.3
STRESS EXPANSION FOR NODE 101			
0.9	9.5	-0.4	0.0 -0.4 -0.0
STRESS EXPANSION FOR NODE 102			
-1.5	9.5	0.7	0.0 -0.4 -0.0
3-D LINE ELEMENT 2	SIZE 0.250	LENGTH 5.500	AREA 0.972
STRESS AT NODES 103, 104	12.2,	12.6	ALLOWABLE 27.3
STRESS EXPANSION FOR NODE 103			
-1.5	9.5	0.7	0.3 -0.4 -0.0
STRESS EXPANSION FOR NODE 104			
-1.8	9.5	0.8	0.3 -0.4 -0.0
3-D LINE ELEMENT 3	SIZE 0.250	LENGTH 37.630	AREA 6.652
STRESS AT NODES 105, 106	11.3,	11.9	ALLOWABLE 27.3
STRESS EXPANSION FOR NODE 105			
0.9	9.2	-0.4	0.0 -0.4 -0.0
STRESS EXPANSION FOR NODE 106			
-1.5	9.2	0.7	0.0 -0.4 -0.0
3-D LINE ELEMENT 4	SIZE 0.250	LENGTH 5.500	AREA 0.972
STRESS AT NODES 107, 108	12.0,	12.3	ALLOWABLE 27.3
STRESS EXPANSION FOR NODE 107			
-1.5	9.2	0.7	0.3 -0.4 -0.0
STRESS EXPANSION FOR NODE 108			
-1.8	9.2	0.8	0.3 -0.4 -0.0

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MJM 8-26-87

AEP \* 150T CRANE \* 50T LOAD \* GIRDER TO END TIE \* WELDS \* SSE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

6	3-D LINE ELEMENT	5	SIZE	0.250	LENGTH	37.630	AREA	6.652	
7	STRESS AT NODES 109,110		5.4,	6.1	ALLOWABLE		27.3		
8	STRESS EXPANSION FOR NODE 109								
9	0.9	-3.3	-0.4	0.0	0.2	-0.0			
10	STRESS EXPANSION FOR NODE 110								
11	-1.5	-3.3	0.7	0.0	0.2	-0.0			
12									
13	3-D LINE ELEMENT	6	SIZE	0.250	LENGTH	5.500	AREA	0.972	
14	STRESS AT NODES 111,112		6.2,	6.5	ALLOWABLE		27.3		
15	STRESS EXPANSION FOR NODE 111								
16	-1.5	-3.3	0.7	0.3	0.2	-0.0			
17	STRESS EXPANSION FOR NODE 112								
18	-1.8	-3.3	0.8	0.3	0.2	-0.0			
19									
20	3-D LINE ELEMENT	7	SIZE	0.250	LENGTH	37.630	AREA	6.652	
21	STRESS AT NODES 113,114		5.7,	6.3	ALLOWABLE		27.3		
22	STRESS EXPANSION FOR NODE 113								
23	0.9	-3.6	-0.4	0.0	0.2	-0.0			
24	STRESS EXPANSION FOR NODE 114								
25	-1.5	-3.6	0.7	0.0	0.2	-0.0			
26									
27	3-D LINE ELEMENT	8	SIZE	0.250	LENGTH	5.500	AREA	0.972	
28	STRESS AT NODES 115,116		6.4,	6.8	ALLOWABLE		27.3		
29	STRESS EXPANSION FOR NODE 115								
30	-1.5	-3.6	0.7	0.3	0.2	-0.0			
31	STRESS EXPANSION FOR NODE 116								
32	-1.8	-3.6	0.8	0.3	0.2	-0.0			
33									
34	3-D LINE ELEMENT	9	SIZE	0.250	LENGTH	58.000	AREA	10.253	
35	STRESS AT NODES 117,118		14.2,	23.2	ALLOWABLE		27.3		
36	STRESS EXPANSION FOR NODE 117								
37	1.0	-11.9	-0.4	-0.7	-0.6	0.1			
38	STRESS EXPANSION FOR NODE 118								
39	1.0	-21.0	-0.4	-0.7	1.1	0.1			
40									
41	3-D LINE ELEMENT	10	SIZE	0.250	LENGTH	3.000	AREA	0.530	
42	STRESS AT NODES 118,119		23.2,	23.2	ALLOWABLE		27.3		
43	STRESS EXPANSION FOR NODE 118								
44	1.0	-21.0	-0.4	-0.7	1.1	0.1			
45	STRESS EXPANSION FOR NODE 119								
46	1.0	-21.0	-0.4	1.0	1.1	-0.1			
47									
48	3-D LINE ELEMENT	11	SIZE	0.250	LENGTH	3.500	AREA	0.619	
49	STRESS AT NODES 104,120		12.6,	13.1	ALLOWABLE		27.3		
50	STRESS EXPANSION FOR NODE 104								
51	-1.8	9.5	0.8	0.3	-0.4	-0.0			
52	STRESS EXPANSION FOR NODE 120								
53	-1.8	9.5	0.8	2.3	-0.4	-0.3			
54									
55	3-D LINE ELEMENT	12	SIZE	0.250	LENGTH	3.500	AREA	0.619	
56	STRESS AT NODES 108,121		12.3,	12.8	ALLOWABLE		27.3		
57	STRESS EXPANSION FOR NODE 108								
58	-1.8	9.2	0.8	0.3	-0.4	-0.0			
59	STRESS EXPANSION FOR NODE 121								
60	-1.8	9.2	0.8	2.3	-0.4	-0.3			

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AEP \* 150T CRANE \* 50T LOAD \* GIRDER TO END TIE \* WELDS \* SSE

\*\*\*\*\* SYSTEM ELEMENT STRESSES \*\*\*\*\*

6	3-D LINE ELEMENT	13	SIZE	0.250	LENGTH	3.500	AREA	0.619	
7	STRESS AT NODES 116,122		6.8,	7.7	ALLOWABLE		27.3		
8	STRESS EXPANSION FOR NODE 116								
9	-1.8	-3.6	0.8	0.3	0.2	-0.0			
10	STRESS EXPANSION FOR NODE 122								
11	-1.8	-3.6	0.8	2.3	0.2	-0.3			
13	3-D LINE ELEMENT	14	SIZE	0.250	LENGTH	3.500	AREA	0.619	
14	STRESS AT NODES 112,123		6.5,	7.4	ALLOWABLE		27.3		
15	STRESS EXPANSION FOR NODE 112								
16	-1.8	-3.3	0.8	0.3	0.2	-0.0			
17	STRESS EXPANSION FOR NODE 123								
18	-1.8	-3.3	0.8	2.3	0.2	-0.3			
19	2-D POINT ELEMENT	15	NODE	51	AREA	1.485	DIAMETER	1.375	
21	STRESS AT NODE		10.4	ALLOWABLE		27.3			
22	STRESS EXPANSION FOR NODE 51								
23	1.0	8.2	-0.4	0.2	0.0	0.0			
25	2-D POINT ELEMENT	16	NODE	52	AREA	1.485	DIAMETER	1.375	
26	STRESS AT NODE		5.4	ALLOWABLE		27.3			
27	STRESS EXPANSION FOR NODE 52								
28	1.0	3.1	-0.4	0.2	0.0	0.0			
30	2-D POINT ELEMENT	17	NODE	53	AREA	1.485	DIAMETER	1.375	
31	STRESS AT NODE		4.3	ALLOWABLE		27.3			
32	STRESS EXPANSION FOR NODE 53								
33	1.0	-2.0	-0.4	0.2	0.0	0.0			
35	2-D POINT ELEMENT	18	NODE	54	AREA	1.485	DIAMETER	1.375	
36	STRESS AT NODE		9.3	ALLOWABLE		27.3			
37	STRESS EXPANSION FOR NODE 54								
38	1.0	-7.1	-0.4	0.2	0.0	0.0			
40	2-D POINT ELEMENT	19	NODE	55	AREA	1.485	DIAMETER	1.375	
41	STRESS AT NODE		14.4	ALLOWABLE		27.3			
42	STRESS EXPANSION FOR NODE 55								
43	1.0	-12.2	-0.4	0.2	0.0	0.0			
45	2-D POINT ELEMENT	20	NODE	56	AREA	1.485	DIAMETER	1.375	
46	STRESS AT NODE		19.5	ALLOWABLE		27.3			
47	STRESS EXPANSION FOR NODE 56								
48	1.0	-17.3	-0.4	0.2	0.0	0.0			
50	FORCE DEFINITION NODE ABSOLUTE ELEMENT 21 NODE 200								
51	FX	57.70 ;	FY	49.50 ;	FZ	2.20			
52	MX	10.50 ;	MY	103.30 ;	MZ	6359.00			

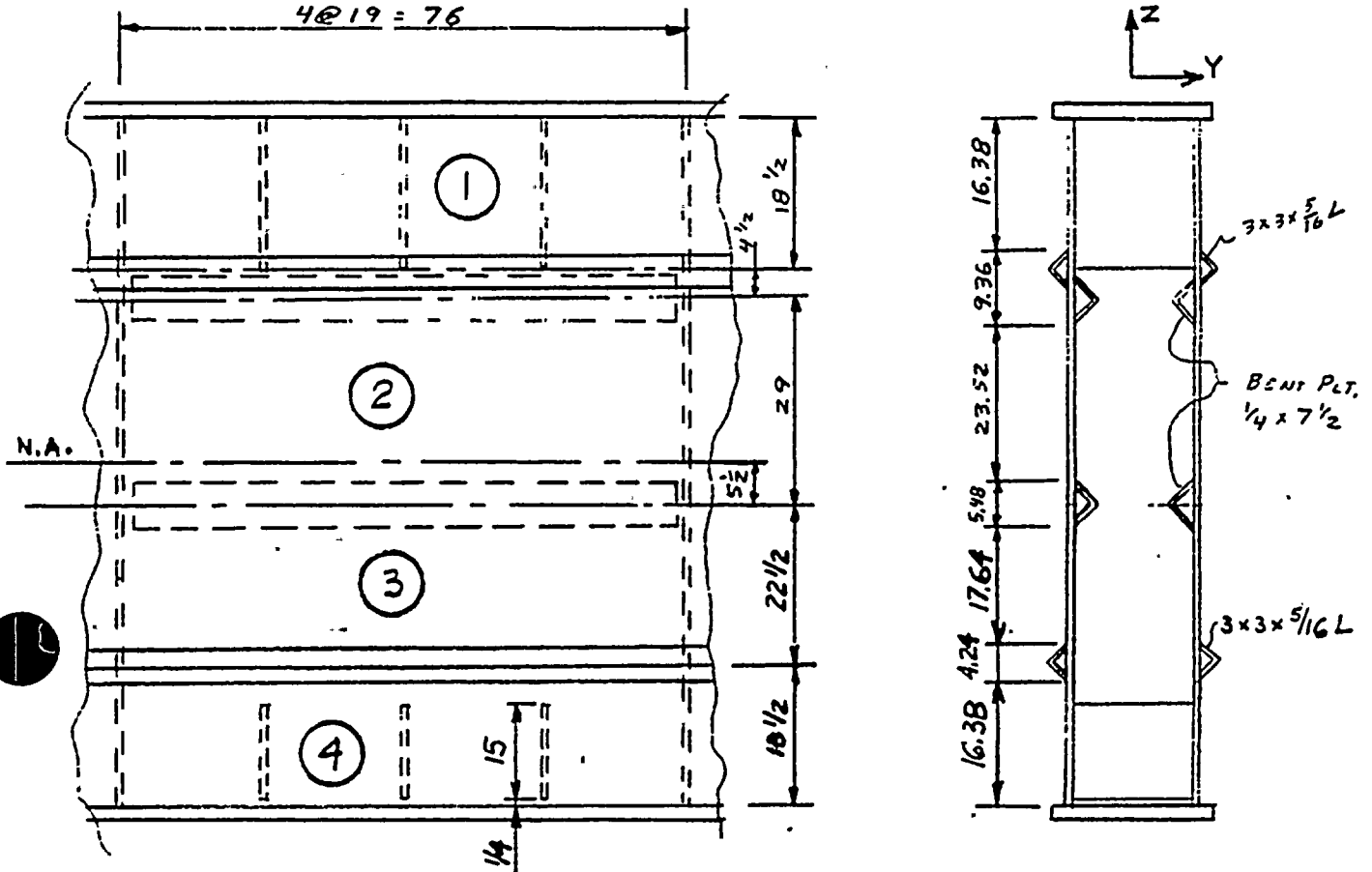
MAXIMUM ABSOLUTE STRESS ON ELEMENT 10 = 23.2 KSI

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WHITING REON. 79604 DATE 8-17-87  
 BY ASZ PAGE 4-93 OF 112  
 MJM 8-25-87

BUCKLING STABILITY OF GIRDER WEB  
 (REF 16 - USS STEEL DESIGN MANUAL -1981, PP. 73-95)



FOR CALCULATIONS ASSUME ALL EDGES ARE SIMPLY SUPPORTED.

GIRDER SECTION PROPERTIES PER PG. 3-13.

MATERIAL: ASTM-A36

$$\sigma_{YMIN} = 36 \text{ KSI}$$

$$E = 29000 \text{ KSI}$$

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WHITING REQ. 79604 DATE 8-17-87  
 BY ASZ PAGE 4-34 OF 112  
 MJM 8-25-87

LONGITUDINAL STIFFENERS CHECK TO DETERMINE THE EFFECTIVENESS OF EACH STIFFENER HAS WHEN DETERMINING THE CRITICAL STRESS OF EACH PANEL.

$$\gamma_1 = \phi + \left(\frac{a}{b_w}\right)^2 \frac{A_s}{b_w t} k_c$$

$$\gamma_2 = \frac{12(1-\nu^2)I_s}{b_w t^3}$$

$\phi$  AND  $k_c$  ARE DEPENDANT ON EACH OTHER AND THE VALUES MUST BE DETERMINED SO THAT THE FOLLOWING IS TRUE. (FIG. 4.13)

$$(k_{c \text{ MAX}} = 64 \text{ AND } \phi_{\text{MAX}} = 11.0)$$

THE EFFECTIVENESS OF THE STIFFENER IS DETERMINED BY THE FOLLOWING

$$\text{RATIO } \beta_{\text{EFF.}} = \frac{k_c}{k_{c \text{ MAX}}} = \frac{k_c}{64}$$

$$\nu = 0.3 \quad a = 76 \text{ IN.} \quad b_w = 93 \text{ IN.} \quad t = 5/16 \text{ IN.} \quad \frac{a}{b_w} = 0.82$$

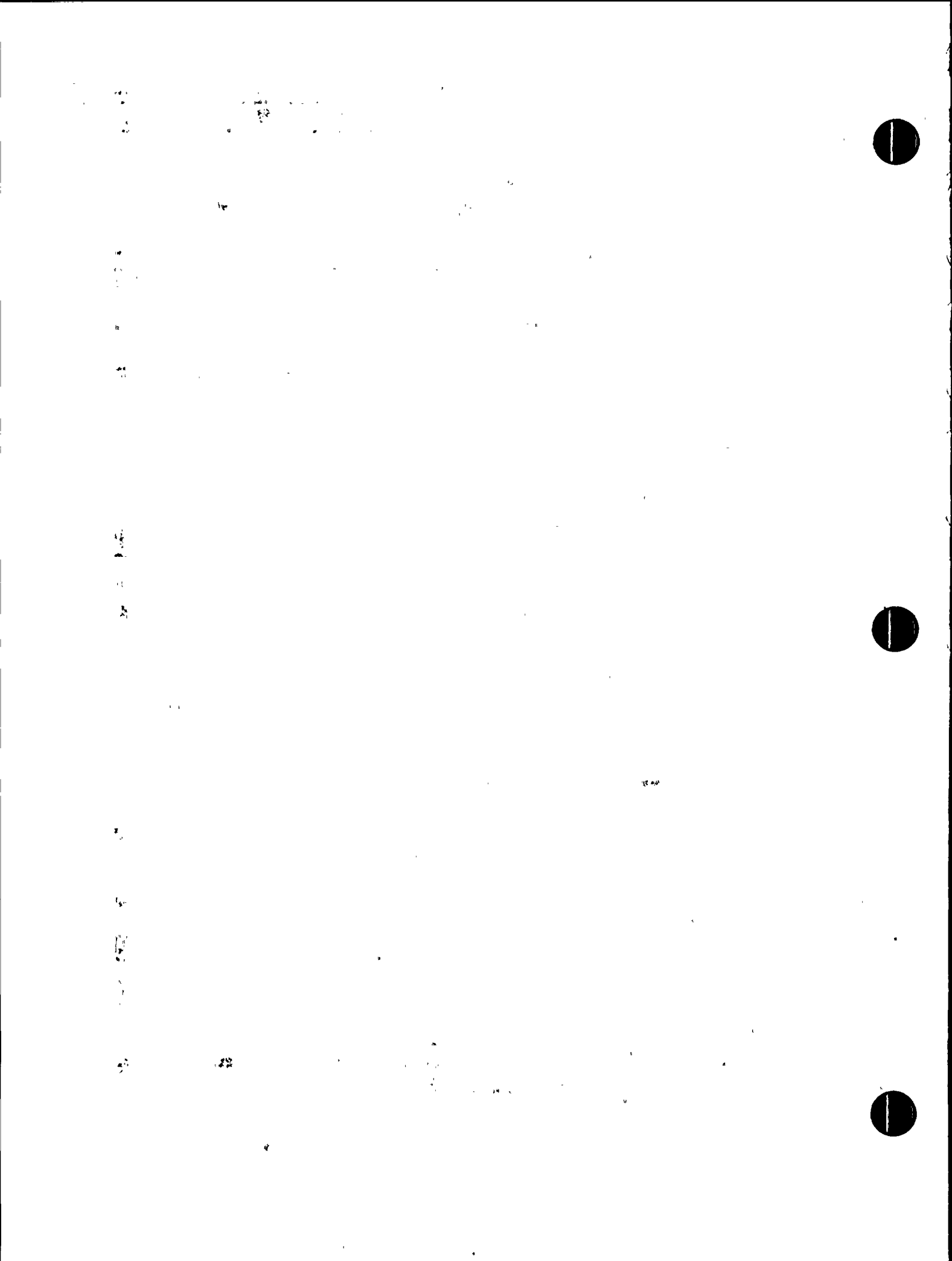
$$\text{EFFECTIVE PLATE WIDTH } e = \frac{1}{2} \frac{6000t}{\sqrt{6y}} = \frac{1}{2} \frac{6000 \times 0.3125}{\sqrt{36000}} = 4.94 \text{ IN.}$$

BOTTOM STIFFENER 3x3x5/16 ANGLE

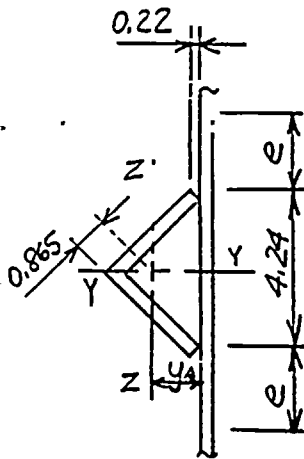
WEB

$$\bar{y}_w = \frac{1}{2} \times \frac{5}{16} = 0.156 \text{ IN.} \quad A_w = (2 \times 4.94 + 4.24) 0.3125 = 4.4 \text{ IN}^2$$

$$I_{zw} = \frac{0.3125^3 (2 \times 4.94 + 4.24)}{12} = 0.04 \text{ IN}^4$$



WHITING REON. 79604 DATE 8-17-87  
 BY ASZ PAGE 4-95 OF 112  
 MJM 8-25-87



ANGLE

$$\bar{y}_A = 3 \cos 45^\circ - \frac{0.865}{\cos 45^\circ} + 0.22 = 1.12 \text{ IN.}$$

$$A_A = 1.78 \text{ IN}^2 \quad r_A = 0.589 \text{ IN.}$$

$$I_{ZA} = r_A^2 A_A = 0.589^2 \times 1.78 = 0.62 \text{ IN}^4$$

TOTAL FOR SECTION

$$\bar{y}_s = \frac{1.78 (1.12 + 0.156)}{6.18} = 0.37 \text{ IN}$$

$$A_s = 4.4 + 1.78 = 6.18 \text{ IN}^2$$

$$I_s = 0.04 + 4.4 \times 0.37^2 + 0.62 + 1.78 (1.28 - 0.37)^2 = 2.73 \text{ IN}^4$$

$$\frac{a}{b_w} = 0.82$$

FROM FIG. 4.13  $\rightarrow \phi = 5.5 \quad k_c = 34.7$

$$\frac{\lambda_1}{\lambda_2} = \frac{5.5 + 0.82^2 \frac{6.18}{93 \times 0.3125} 34.7}{\frac{10.9 \times 2.73}{93 \times 0.3125^3}} = 0.998 \leq 1.0$$

OK

$\therefore$  STIFFENER EFFECTIVENESS  $\beta_{\text{EFF BOTTOM}} = \frac{34.7}{64} = 0.54 = 54\%$

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1962

1962

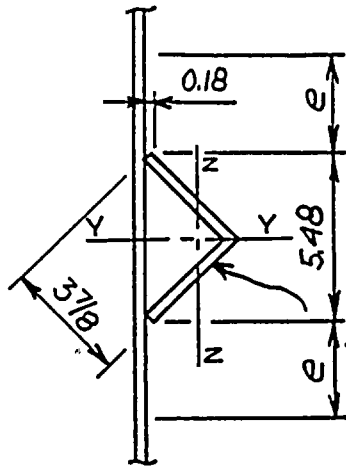
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WHITING REQ. 79604 DATE 8-17-27  
 BY ASZ PAGE 4-96 OF 112  
 MJM 8-25-87

CENTER STIFFENER 1/4 x 7 1/2 BENT PL

WEB



$$\bar{y}_w = \frac{1}{2} 0.3125 = 0.156 \text{ IN.}$$

$$A_w = (2 \times 4.94 + 5.48) 0.3125 = 4.8 \text{ IN}^2$$

$$I_w = \frac{0.3125^3 (2 \times 4.94 + 5.48)}{12} = 0.04 \text{ IN}^4$$

BENT PL 1/4 x 7 1/2

$$a = 3 7/8 \text{ IN. } t = \frac{1}{4} \text{ IN. } b = a - t = 3 5/8 \text{ IN.}$$

$$\bar{y}_b = 3.875 \cos 45^\circ - \frac{3.875^2 + 3.875 \times 0.25 - 0.25^2}{2(2 \times 3.875 - 0.25) \cos 45^\circ}$$

$$+ 0.18 = 1.42 \text{ IN.}$$

$$y = 1.5$$

$$I_{zb} = \frac{A_b}{12} [7(a^2 + b^2) - 12y^2] - 2ab^2(a - b) = 1.12 \text{ IN}^4$$

$$A_b = 0.25 \times 7.5 = 1.875 \text{ IN}^2$$

TOTAL FOR SECTION

$$\bar{y}_s = \frac{1.875 (1.42 + 0.156)}{6.675} = 0.44 \text{ IN}$$

$$A_s = 4.8 + 1.875 = 6.675 \text{ IN}^2$$

$$I_s = 0.04 + 4.8 \times 0.44^2 + 1.12 + 1.875 (1.58 - 0.44)^2 = 4.5 \text{ IN}^4$$

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WHITING REQ. 79604 DATE 8-18-87  
 BY ASZ PAGE 4-97 OF 112  
 MJM 8-25-87

$$\frac{a}{b_w} = 0.82$$

FROM FIG 4.13  $\rightarrow \phi = 9.0 \quad k_c = 53$

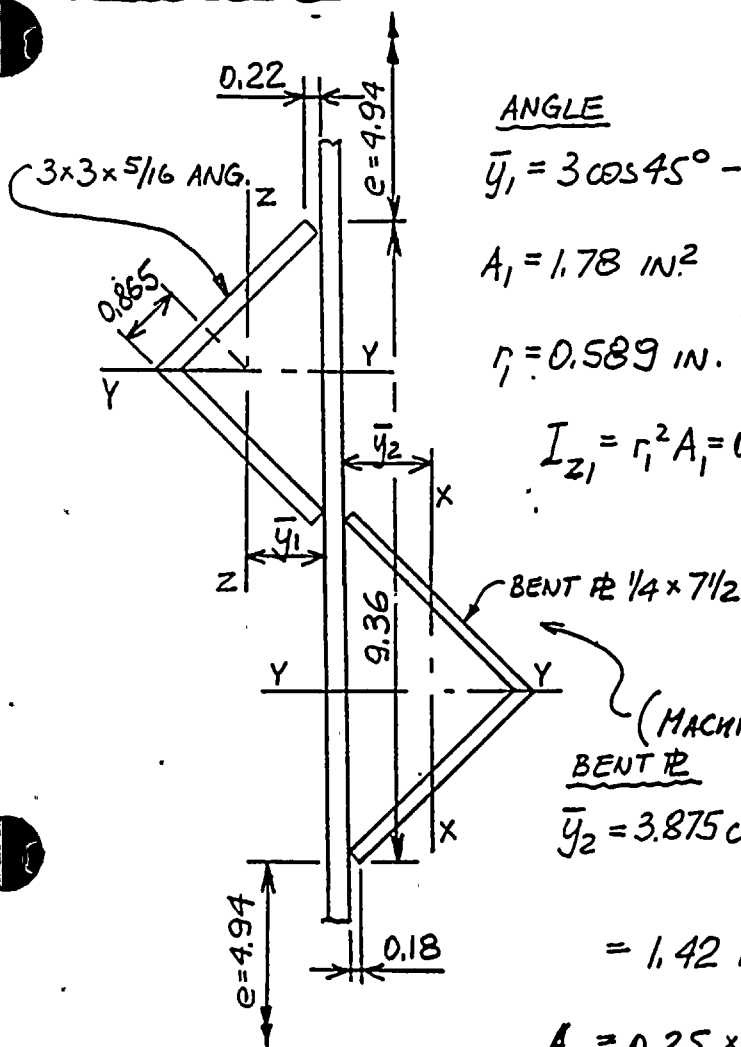
$$\frac{\gamma_1}{\gamma_2} = \frac{9.0 + 0.82^2 \frac{6.675}{93 \times 0.3125} 53}{\frac{10.9 \times 4.5}{93 \times 0.3125^3}} = 0.99 \leq 1.0$$

OK

• STIFFENER  
 • EFFECTIVENESS  $\beta_{EFF. CENTER} = \frac{53}{64} = 0.83 = 83\%$

WHITING REQN. 79604 DATE 8-18-87  
 BY ASZ PAGE 4-98 OF 112  
 MJM 8-25-87

TOP STIFFENER - ANGLE AND BENT P.



ANGLE

$$\bar{y}_1 = 3 \cos 45^\circ - \frac{0.865}{\cos 45^\circ} + 0.22 = 1.12 \text{ IN.}$$

$$A_1 = 1.78 \text{ IN}^2$$

$$r_1 = 0.589 \text{ IN.}$$

$$I_{z_1} = r_1^2 A_1 = 0.589^2 \times 1.78 = 0.62 \text{ IN}^4$$

(MACHINERY'S HANDBOOK, 22/ED, PG. 266)

BENT P.

$$\bar{y}_2 = 3.875 \cos 45^\circ - \frac{3.875^2 + 3.875 \times 0.25 - 0.25^2}{2(2 \times 3.875 - 0.25) \cos 45^\circ} + 0.18$$

$\underbrace{\hspace{10em}}_y$

$$= 1.42 \text{ IN.}$$

$$A_2 = 0.25 \times 7.5 = 1.875 \text{ IN}^2$$

$$I_{z_2} = \frac{A_2}{12} [7(a^2 + b^2) - 12y^2] - 2ab^2(a-b)$$

$$a = 3\frac{7}{8} \text{ IN.} \quad t = 0.25 \text{ IN} \quad b = a - t = 3\frac{5}{8} \text{ IN.} \quad y = 1.50 \text{ IN.}$$

$$I_{z_2} = 1.12 \text{ IN}^4$$

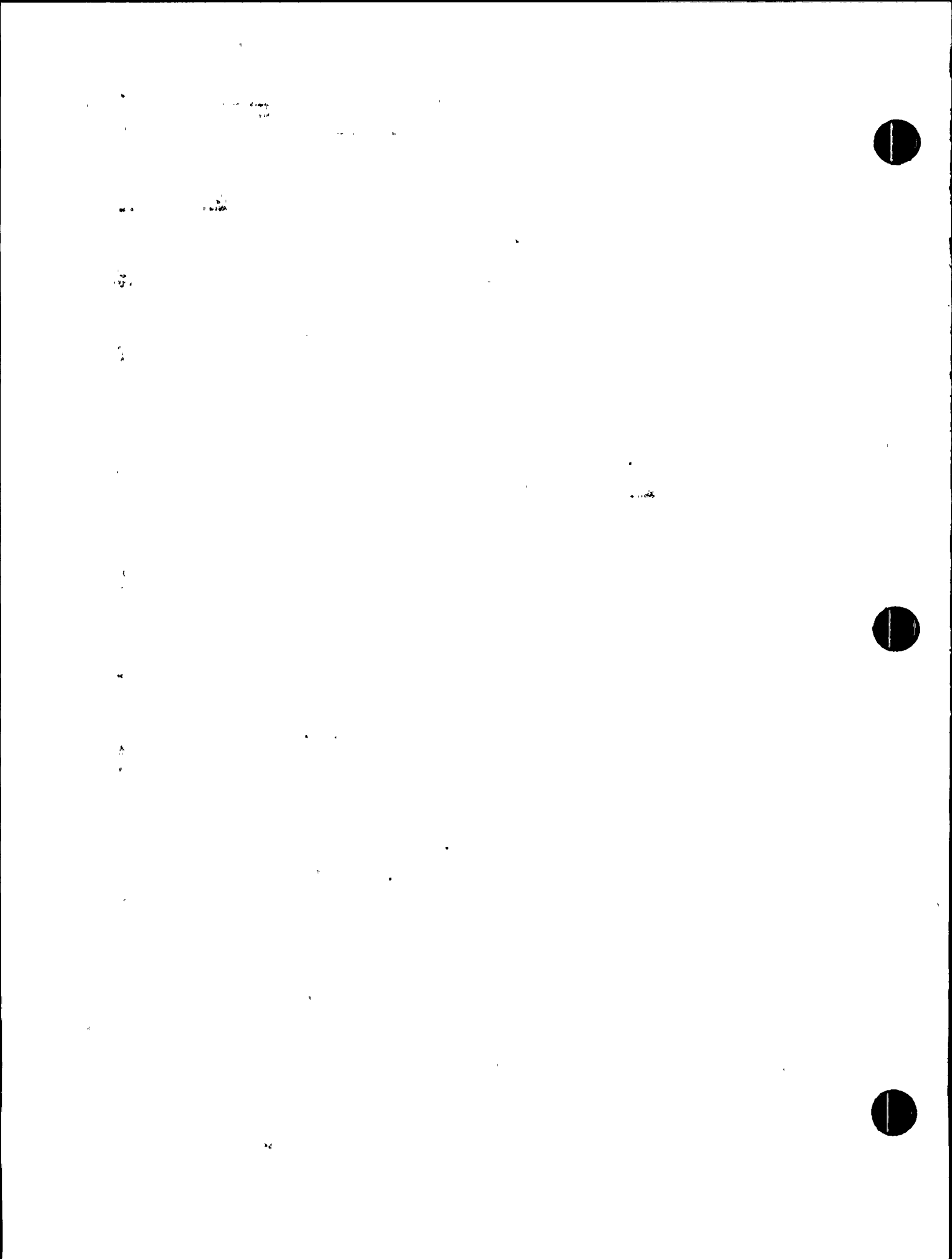
WEB

$$\bar{y}_w = \frac{1}{2} \times 0.3125 = 0.156 \text{ IN}$$

$$A_w = (2 \times 4.94 + 9.36) \times 0.3125 = 6.01 \text{ IN}^2$$

$$I_w = \frac{0.3125^3 \times 19.24}{12} = 0.05 \text{ IN}^4$$





WHITING REQ. 79604 DATE 8-18-87  
 BY ASZ PAGE 4-99 OF 112  
 MJM 8-25-87

TOTAL FOR SECTION

$$\bar{y} = \frac{1.78(1.12 + 0.156) - 1.88(1.41 + 0.156)}{9.67} = -0.07 \text{ IN.}$$

$$A_s = 6.01 + 1.78 + 1.88 = 9.67 \text{ IN}^2$$

$$I_x = I_s = 0.05 + 6.01(0.07)^2 + 0.62 + 1.78(1.28 - 0.07)^2 + 1.12 \\ + 1.875(1.57 - 0.07)^2 = 8.64 \text{ IN}^4$$

$$\frac{\sigma}{b_w} = 0.82$$

FROM FIG. 4.13  $\rightarrow \phi = 11.0 \quad k_c = 64$

$$\frac{\lambda_1}{\lambda_2} = \frac{11.0 + 0.82^2 \frac{9.67}{93 \times 0.3125} 64}{\frac{10.9 \times 8.64}{93 \times 0.3125^3}} = 0.76$$

$\therefore$  STIFFENER EFFECTIVENESS  $\beta_{\text{EFF. TOP}} = \frac{64}{64} = 1.0 = 100\%$

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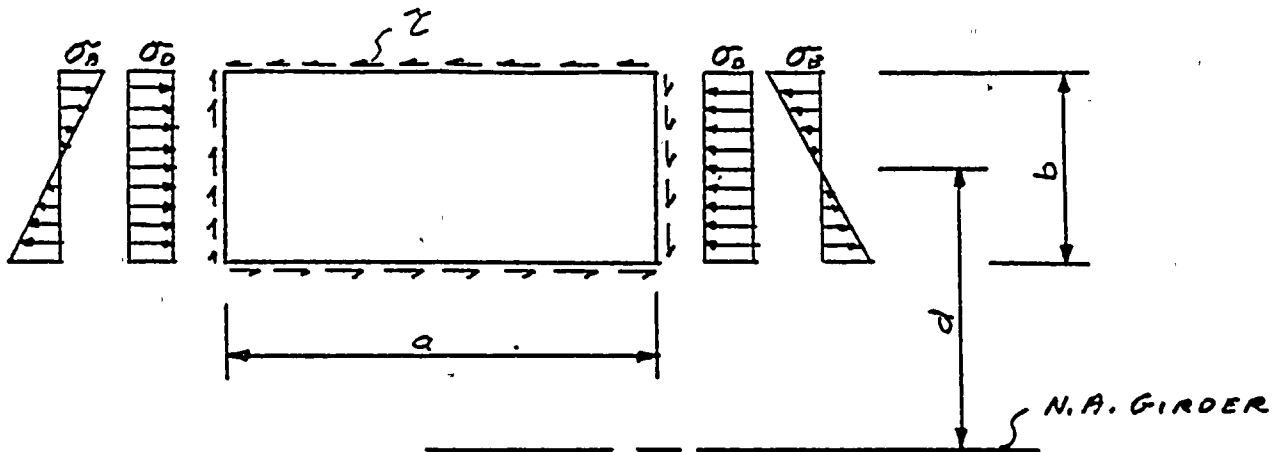
117

118



WHITING REQN. 79604 DATE 8-18-87  
 BY ASZ PAGE 4-100 OF 112  
 MJM 8-25-87

WEB PANEL STABILITY CHECK



FOR BUCKLING STABILITY THE FOLLOWING CRITERIA MUST BE MET

$$\frac{\sigma_D}{\sigma_{Dcr}} + \left[ \frac{\sigma_B}{\sigma_{Bcr}} \right]^2 + \left[ \frac{\tau}{\tau_{cr}} \right]^2 = R \leq \frac{1}{\text{D.F.}} = \frac{\sigma_{all}}{\sigma_{min}}$$

DESIGN FACTOR

THE CRITICAL STRESS =  $k \frac{\pi^2 E J^2}{12(1-N^2)b^2}$       $\left\{ \begin{array}{l} J = 5/16 \text{ in} \\ E = 29,000 \text{ KSI} \\ N = .3 \end{array} \right.$

SET  $\psi = \frac{\pi^2 E J^2}{12(1-N^2)b^2} = \frac{\pi^2 (29000) (5/16)^2}{12(1-.3^2)b^2} = \frac{2560}{b^2}$

CRITICAL STRESS =  $\beta_{EFF} k \psi$  KSI, WHERE  $\beta_{EFF}$  IS THE EFFECTIVENESS RATIO OF THE WEB STIFFENER

IF THE ELASTIC CRITICAL STRESS EXCEEDS THE MINIMUM YIELD STRESS, THEN THE CRITICAL STRESS IS SET TO EQUAL THE YIELD STRESS

$$\sigma_{cr \text{ MAX}} = \sigma_{min} = 36.0 \text{ KSI}$$

$$\tau_{cr \text{ MAX}} = \tau_{min} = \frac{\sigma_{min}}{\sqrt{3}} = \frac{36}{\sqrt{3}} = 20.8 \text{ KSI}$$



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WHITING REQ. 79604 DATE 8-18-87  
 BY ASZ PAGE 4-101 OF 112  
 MJM 8-25-87

DIRECT STRESS

$$\begin{aligned} G_D &= \frac{F_x}{A} + \frac{M_y d}{I_y} + \frac{M_z C_w}{I_z} = \frac{F_x}{110} + \frac{M_y d}{158490} + \frac{M_z \frac{17.3125}{2}}{6285} \\ &= \frac{F_x}{110} + \frac{M_y d}{158490} + \frac{M_z}{726} \end{aligned}$$

BENDING STRESS

$$G_B = \frac{M_y (b/2)}{158490} = \frac{M_y b}{316980}$$

SHEAR STRESS

$$\begin{aligned} \tau &= \frac{F_z}{2 A_{web}} + \frac{M_x}{2 A_{fl} t} = \frac{F_z}{2 \times 93 \times 0.3125} + \frac{M_x}{2 \times 17.3125 \times 94.25 \times 0.3125} \\ &= \frac{F_z}{58} + \frac{M_x}{1020} \end{aligned}$$

THE ALLOWABLE STABILITY RATIO IS 1.0 DIVIDED BY THE DESIGN FACTOR (OR THE ALLOWABLE STRESS DIVIDED BY THE MINIMUM YIELD STRENGTH OF THE WEB)

$$\therefore \text{OBE } R_{ALL} = \frac{1}{1.5} = 0.667$$

$$\text{SSE } R_{ALL} = \frac{1}{1.1} = 0.909$$

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WHITING REQN. 79604 DATE 8-11-87  
 BY ASZ PAGE 4-111R OF 112  
 MJM 8-26-87  
 REV. ASZ 9-14-87 04/19/14/87

## ROPE

MAIN HOIST ROPE: 1 1/2 DIA. PYTHON 10 FIG V, XIPS RIGHT HAND

REGULAR LAY ALL STEEL ROPE, TWO-6 PART SYSTEMS

MIN. BREAKING STRENGTH  $P_{BR} = 160.9 T$

ALLOWABLE ROPE LOAD  $P_{ALL} = \frac{P_{BR} f_p n m}{f}$

PROP. LMT. FACTOR  $f_p = 0.6$

OBE FACTOR  $f = 1.5$

NO. OF ROPE PARTS  $n = 6$

SSE FACTOR  $f = 1.1$

NO. OF ROPE SYSTEMS  $m = 2$

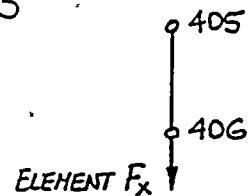
$$\underline{OBE} \quad P_{ALL} = \frac{160.9 \times 0.6 \times 6 \times 2}{1.5} = 772.3 T$$

$$\underline{SSE} \quad P_{ALL} = \frac{160.9 \times 0.6 \times 6 \times 2}{1.1} = 1053 T$$

MAX. ROPE LOAD PER TABLES B 61 AND B 63

$$\underline{OBE} \quad F_x = 367.3 \text{ KIP} = 183.6 T$$

$$\underline{SSE} \quad F_x = 583.7 \text{ KIP} = 291.8 T$$



$$\text{RATIO } R = \frac{F_x}{P_{ALL}} \leq 1.0$$

$$\underline{OBE} \quad R = \frac{183.6}{772.3} = 0.24$$

$$\underline{SSE} \quad R = \frac{291.8}{1053} = 0.28$$





WHITING REQN. 79604 DATE 2-11-27  
 BY ASZ PAGE 4-112 OF 112  
 MJM 8-26-87

DYNAMIC  
LOAD FACTOR  $DLF = \frac{F_x}{P_S}$

$$DLF' = \frac{F_x}{P_S'}$$

STATIC LOAD  $P_S = W + P_L$

TROLLEY BLOCK WEIGHT  $W = 20000 \text{ LBS} = 10T$

SEISMIC  
LIVE LOAD  $P_L = 50T$

$$P_S = 10 + 50 = 60T$$

STATIC LOAD  $P_S' = W + P_L'$

RATED  
LIVE LOAD  $P_L' = 150T$

$$P_S' = 10 + 150 = 160T$$

OBE  $DLF = \frac{183.6}{60} = 3.1$

$$DLF' = \frac{183.6}{160} = 1.1$$

SSE  $DLF = \frac{291.8}{60} = 4.9$

$$DLF' = \frac{291.8}{160} = 1.8$$

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WHITING REQ. 79604 DATE 2-8-27  
BY ASZ PAGE 4-102 OF 112  
MJM 8-25-97

FOR PANELS 1 AND 2, WHICH ARE ABOVE THE GIRDER'S NEUTRAL AXIS, THE  $M_y$  MOMENT IS TAKEN AS THE SUM OF STATIC AND DYNAMIC. FOR PANELS 3 AND 4, WHICH ARE BELOW THE NEUTRAL AXIS, THE  $M_y$  IS TAKEN AS THE DIFFERENCE OF STATIC AND DYNAMIC.

FOR GIRDER STABILITY, IT HAS BEEN DETERMINED THAT THE DOMINANT LOAD CONDITION WOULD BE WITH THE LOAD IN THE DOWN POSITION (THIS PRODUCES THE MAXIMUM GIRDER STRESS). THE LOADINGS WERE TAKEN FROM TABLES B65 AND B66. IT IS FOUND THAT THE STABILITY RATIO IS MAXIMUM FOR THE TROLLEY AT MID.

WHITING REQ. 79604 DATE 8-18-87  
 BY ASZ PAGE 4-103 OF 112  
 MJM 8-25-87

TABLE 4-6

	PANEL 1	PANEL 2	PANEL 3	PANEL 4
<i>a</i>	19	76	76	19
<i>b</i>	16.38	23.52	17.64	16.38
<i>d</i>	38.31	9.0	-17.06	-38.31
$\psi$	9.54	4.63	8.23	9.54
$k_D$	4.0	4.0	4.0	4.0
$k_B$	23.9	23.9	23.9	23.9
$k_S$	8.3	5.8	5.6	8.3
$\beta_{EFF}$	1.0	0.83	0.54	0.54
$G_{DCR}$	36.0	15.4	17.8	20.6
$G_{BCR}$	36.0	36.0	36.0	36.0
$T_{CR}$	20.8	20.8	20.8	20.8
<u>OBE</u>				
$R_{MID-A}$	0.590	0.556	0.264	0.246
$R_{MID-B}$	0.594	0.597	0.315	0.297
<u>SSE</u>				
$R_{MID-A}$	0.864	0.816	0.564	0.712
$R_{MID-B}$	0.870	0.876	0.627	0.764

OBE  $R_{MAX} = 0.597 < 0.667 \therefore OK$

SSE  $R_{MAX} = 0.876 < 0.909 \therefore OK$



WHITING REQ. 79604 DATE 5-26-87  
 BY ASZ PAGE 4-105 OF 112  
 MJM 8-26-87

GIRDER END

(REF. DWG. ON PG. 4-37, SECTION A-A)

MTRL: ASTM-A36

$$S_{YMIN} = 36 \text{ KSI}$$

ALLOWABLES

OBE  $\tau_{ALL} = \frac{S_{YMIN}}{1.5} 0.6 = 14.4 \text{ KSI}$   $\tau_{W,ALL} = 20.0 \text{ KSI}$   
 (REF. PG. 4-38)

SSE  $\tau_{ALL} = \frac{S_{YMIN}}{1.1} 0.6 = 19.6 \text{ KSI}$   $\tau_{W,ALL} = 27.3 \text{ KSI}$

MAX. LOADS (REF. PG. 4-40). OBE  $F_2 = 250.2 \text{ KIP}$

SSE  $F_2 = 339.4 \text{ KIP}$

OTHER LOADS ARE TRANSFERRED THROUGH BOLTED CONNECTION

WEB SHEAR STRESS  $\tau = \frac{F_2 Q}{I_{xx} t}$

OBE  $\tau = \frac{250.2 [21 \times 1.25 (18.53 - 0.625) + 2 \times 0.3125 (18.53 - 1.25)^2 0.5]}{22150 \times 0.3125 \times 2} = 10.2 \text{ KSI}$

SSE  $\tau = \frac{339.4 \times 563.3}{22150 \times 0.3125 \times 2} = 13.8 \text{ KSI}$

UPPER WELDS SHEAR STRESS

OBE  $\tau_N = \frac{250.2 (21 \times 1.25) (18.53 - 0.625)}{2 \times 0.3125 \times 0.707 \times 22150} = 12.0 \text{ KSI}$

SSE  $\tau_N = \frac{339.4 \times 21 \times 1.25 (18.53 - 0.625)}{2 \times 0.3125 \times 0.707 \times 22150} = 16.3 \text{ KSI}$

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WHITING REQ. 79604 DATE 6-26-87  
 BY ASZ PAGE 4-106 OF 112  
 MJM 8-26-87

LOWER WELDS  
 SHEAR STRESS

$$\underline{OBE} \tau_w = \frac{250.2 (2 \times 6.75) (45.375 - 18.53 - 1.47)}{2(0.25) 0.707 \times 22150}$$

$$= 10.9 \text{ ksi}$$

$$\underline{SSE} \tau_w = \frac{339.4 \times 2 \times 6.75 (45.375 - 18.53 - 1.47)}{2(0.25) 0.707 \times 22150} = 14.8 \text{ ksi}$$

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# GIRDER END

## SECTION PROPERTIES

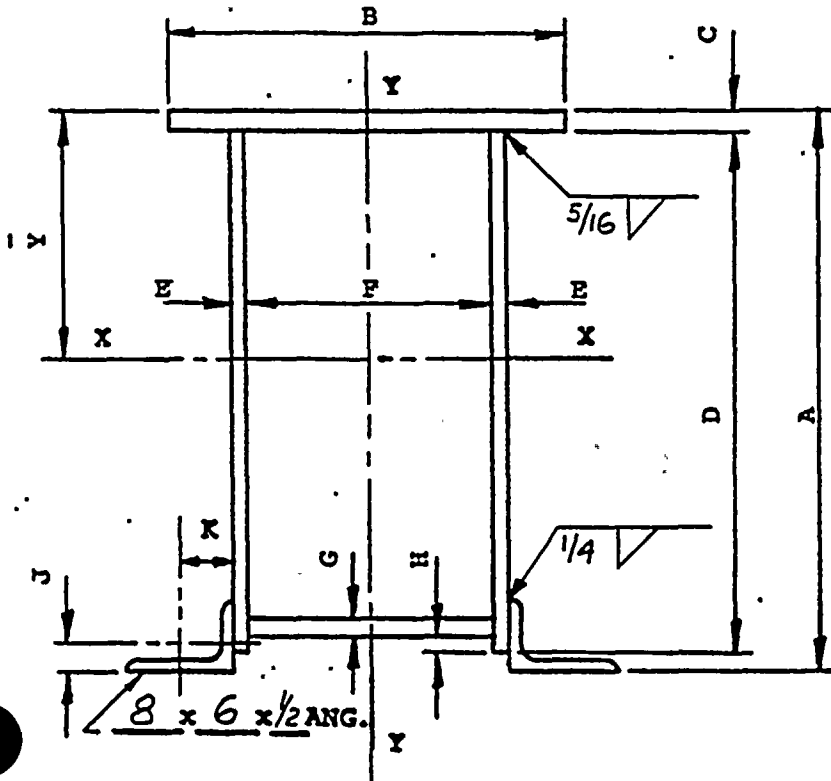
### SECTION "AA"

PROGRAM 117 PROGRAM ID 1-A-2-06 (046)

WHITING REQ# 79604 DATE 6-25-87

BY ASZ PAGE 4-107 OF 112

MJM 8-26-87



### GIVEN DATA

1.	<u>45.375</u>	= DIMENSION A (IN.)-----	45.3750
2.	<u>21</u>	= DIMENSION B (IN.)-----	21.0000
3.	<u>1.25</u>	= DIMENSION C (IN.)-----	1.2500
4.	<u>43.875</u>	= DIMENSION D (IN.)-----	43.8750
5.	<u>0.3125</u>	= DIMENSION E (IN.)-----	0.3125
6.	<u>17</u>	= DIMENSION F (IN.)-----	17.0000
7.	<u>0</u>	= DIMENSION G (IN.)-----	0.0000
8.	<u>0</u>	= DIMENSION H (IN.)-----	0.0000
9.	<u>1.47</u>	= DIMENSION J (IN.) (ȳ OF ANGLE)-----	1.4700
10.	<u>2.47</u>	= DIMENSION K (IN.) (x̄ OF ANGLE)-----	2.4700
11.	<u>6.75</u>	= AREA OF ANGLE (IN. <sup>2</sup> )-----	6.7500
12.	<u>44.3</u>	= MOMENT OF INERTIA OF ANGLE (IN <sup>4</sup> ) (ABT. VERT)-----	44.3000
13.	<u>21.7</u>	= MOMENT OF INERTIA OF ANGLE (IN <sup>4</sup> ) (ABT. HORZ)-----	21.7000

### COMPUTED DATA

I <sub>y-y</sub> (IN. <sup>4</sup> ) (MOMENT OF INERTIA OF SECTION)-----	4826.7298
ȳ (IN.)-----	18.5340
I <sub>x-x</sub> (IN. <sup>4</sup> ) (MOMENT OF INERTIA OF SECTION)-----	22148.6073

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WHITING REQN. 79604 DATE 8-18-87BY MJM PAGE 4-108 OF 112

ASZ. 8/26/87

## TROLLEY WHEEL LOADS

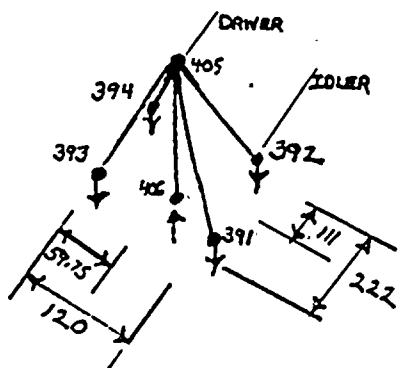
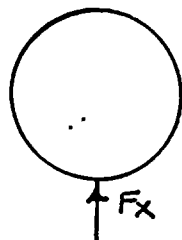
THE TROLLEY REACTIONS ARE SUMMARIZED  
IN TABLES B 57 THROUGH B 60  
WITH THE FORCES IN THE ELEMENT  
COORDINATE SYSTEM

ELEMENT  $F_x$   
(GLOBAL  $F_z$ )

VERTICAL WHEEL LOAD

THE MAXIMUM  $F_x$  IS TAKEN FROM  
TABLE B 57 FOR THE OBE &  
TABLE B 59 FOR THE SSE  
DIRECTLY.

THE MINIMUM  $F_x$  IS TAKEN FROM  
TABLE B 58 FOR THE OBE &  
TABLE B 60 FOR THE SSE  
EXCEPT FOR THE ROPE DOWN  
CASES WHERE THE ROPE IS  
OBSERVED TO BE IN COMPRESSION  
WHICH CAN NOT BE. FOR THESE  
CASES THE PROPORTIONATE AMOUNT  
OF ROPE UPKICK IS DEDUCTED



LOADS AT NODES 391, 392

$$F_{x \text{ MIN}} = F_{x \text{ TABLE}} - F_{\text{ROPE UP}} \left( \frac{111}{222} \right) \left( \frac{59.75}{120} \right)$$

$$= F_{x \text{ TABLE}} - .249 F_{\text{ROPE UP}}$$

LOADS AT NODES 393, 394

$$F_{x \text{ MIN}} = F_{x \text{ TABLE}} - F_{\text{ROPE UP}} \left( \frac{111}{222} \right) \left( \frac{60.25}{120} \right)$$

$$= F_{x \text{ TABLE}} - .251 F_{\text{ROPE UP}}$$

$F_{\text{ROPE UP}}$  IS TAKEN FROM  
TABLE B 62 FOR THE OBE &  
TABLE B 64 FOR THE SSE

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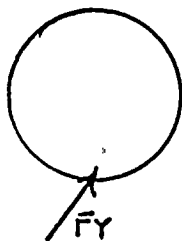
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WHITING, REQN. 79604 DATE 8-13-37  
 BY M J M PAGE 4-10<sup>2</sup> OF 112  
 ASZ 8-26-87

ELEMENT  $F_y$   
 (GLOBAL  $F_y$ )

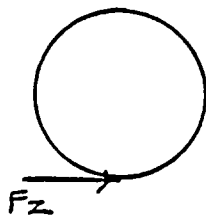


LOADS PERPENDICULAR GIRDER

THE MAXIMUM  $F_y$  IS TAKEN FROM  
 TABLE B 57 FOR THE OBE &  
 TABLE B 59 FOR THE SSE  
 DIRECTLY

THE MINIMUM  $F_y$  IS THE NEGATIVE  
 OF THE MAXIMUM CONSIDERING  
 COMPLETE REVERSAL

ELEMENT  $F_z$   
 (GLOBAL  $F_x$ )



LOADS PARALLEL GIRDER

THE MAXIMUM  $F_z$  IS TAKEN FROM  
 TABLE B 57 FOR THE OBE &  
 TABLE B 59 FOR THE SSE

OR IS THE MAXIMUM  $F_x$  (VERTICAL)  
 TIMES THE COEFFICIENT OF  
 FRICTION (.25) IF THIS IS LESS

THE MINIMUM  $F_z$  IS THE NEGATIVE  
 OF THE MAXIMUM CONSIDERING  
 COMPLETE REVERSAL

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WHITING REQ. 79604 DATE 8-18-87  
 BY MJM PAGE 4-110 OF 112  
 ASZ 8-26-87.

### SUMMARY OF TROLLEY WHEEL LOADS

TABLE 4-7 OBE

TROLLEY	AXLE	F <sub>X</sub> MAX KIP	F <sub>X</sub> MIN KIP	F <sub>Y</sub> MAX KIP	F <sub>Z</sub> MAX KIP
MID	DRIVER	119.1	6.4	25.7	11.9
	IDLER	161.7	20.3	25.7	-
1/4	DRIVER	131.0	.1	29.8	11.9
	IDLER	161.8	25.3	29.8	-
END	DRIVER	120.1	-4.6	39.4	14.1
	IDLER	145.9	25.4	39.4	-

TABLE 4-8 SSE

TROLLEY	AXLE	F <sub>X</sub> MAX KIP	F <sub>X</sub> MIN KIP	F <sub>Y</sub> MAX KIP	F <sub>Z</sub> MAX KIP
MID	DRIVER	182.4	-2.6	36.5	21.9
	IDLER	237.4	-1.4	36.5	-
1/4	DRIVER	193.0	-2.5	43.0	22.5
	IDLER	226.5	16.8	43.0	-
END	DRIVER	173.6	-16.8	56.8	25.6
	IDLER	197.5	13.9	56.8	-

WHITING REQN. 79604 DATE 8-21-87  
 BY MJM PAGE A1 A1 OF 22  
*0224 8-27-87*  
 REV. MJM 9-14-87 *0224 9-15-87*

### APPENDIX A

This appendix summarizes the amplified response spectra and the modal response of the crane.

Page	Table	Title
A-2	A1	Response Spectrum OBE
A-3	A2	Response Spectrum SSE
A-4	A3	Freq & MC, Mid, 50T, UP, OBE
A-5	A4	Freq & MC, Mid, 50T, UP, SSE
A-6	A5	Freq & MC, Mid, 50T, DN, OBE
A-7	A6	Freq & MC, Mid, 50T, DN, SSE
A-8	A7	Freq & MC, 1/4, 50T, UP, OBE
A-9	A8	Freq & MC, 1/4, 50T, UP, SSE
A-10	A9	Freq & MC, 1/4, 50T, DN, OBE
A-11	A10	Freq & MC, 1/4, 50T, DN, SSE
A-12	A11	Freq & MC, End, 50T, UP, OBE
A-13	A12	Freq & MC, End, 50T, UP, SSE
A-14	A-13	Freq & MC, End, 50T, DN, OBE
A-15	A14	Freq & MC, End, 50T, DN, SSE
A-16	A15	Freq & MC, Mid, No Load, OBE
A-17	A16	Freq & MC, Mid, No Load, SSE
A-18	A17	Freq & MC, 1/4, No Load, OBE
A-19	A18	Freq & MC, 1/4, No Load, SSE
A-20	A19	Freq & MC, End, No Load, OBE
A-21	A20	Freq & MC, End, No Load, SSE
A-22	A21	Summary of Computer Runs

MC is the Mode coefficient

(Ref 1, p 2.12.1-2.12.4)

$$MC = \frac{\delta_i S_{ai}}{(2\pi F_i)^2} = \frac{\{d\}_i}{\{\psi\}_i}$$

Where  $S_{ai}$  = modal spectrum acceleration for  $i$ th mode  
 $F_i$  = frequency of  $i$ th mode (eigen value)  
 $\{d\}_i$  = maximum modal displacement vector for  $i$ th mode  
 $\{\psi\}_i$  = normalized modal displacement vector for  $i$ th mode (eigen vector)  
 $\delta_i$  = participation factor for  $i$ th mode  
 $= \{\psi\}_i^T [M] \{D\}$

where  $[\psi]$  is the square matrix containing all mode shape vectors such that the  $i$ th column is the mode shape vector for the  $i$ th mode =  $\{\psi\}_i$   
 $[M]$  = reduced mass matrix  
 $\{D\}$  = unit vector describing excitation

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WHITING REQN. 79604 DATE 6-8-87  
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*077 8.25.87*

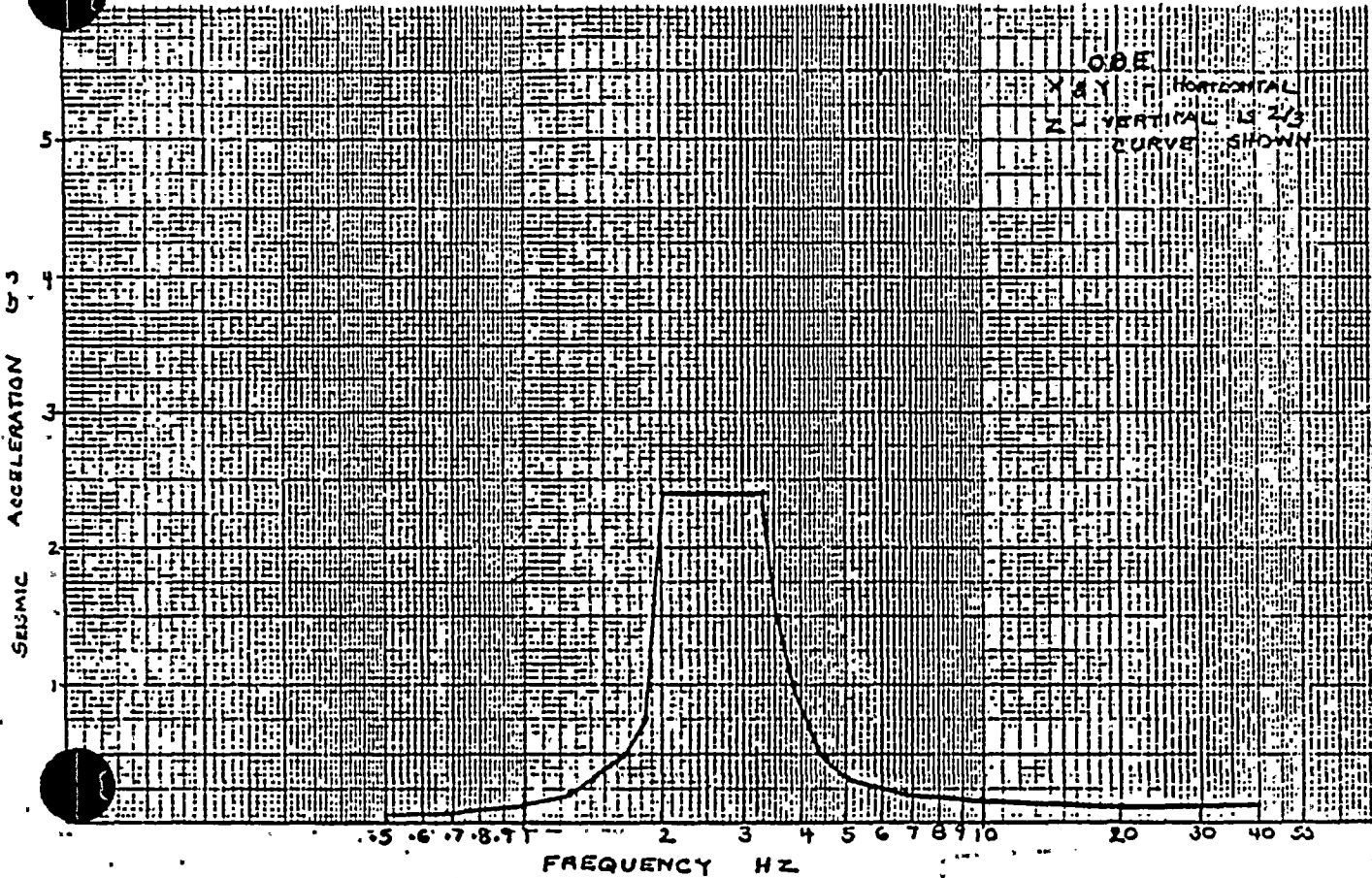


TABLE A-1  
OBE

POINT	FREQ HZ	X OR Y		Z		POINT	FREQ HZ	X OR Y		Z	
		G's	in/sec <sup>2</sup>	G's	in/sec <sup>2</sup>			G's	in/sec <sup>2</sup>	G's	in/sec <sup>2</sup>
1	0.001	.001	.4	.001	.4	11	3.57	1.5	579.6	1.0	386.4
2	.50	.07	27.0	.05	19.3	12	3.85	1.0	386.4	.67	258.9
3	.67	.09	34.8	.06	23.2	13	4.17	.68	262.8	.45	173.9
4	1.0	.15	58.0	.10	38.6	14	4.45	.47	181.6	.31	119.8
5	1.25	.23	88.9	.15	58.0	15	5.0	.34	131.4	.23	88.9
6	1.42	.32	123.6	.21	81.1	16	6.67	.22	85.0	.15	58.0
7	1.67	.50	193.2	.33	127.5	17	10.	.17	65.7	.11	42.5
8	1.81	.75	289.8	.50	193.2	18	20.	.13	50.2	.09	34.8
9	2.0	2.4	927.4	1.6	618.2	19	33.	.12	46.4	.08	30.9
10	3.33	2.4	927.4	1.6	618.2	20	50.	.12	46.4	.08	30.9

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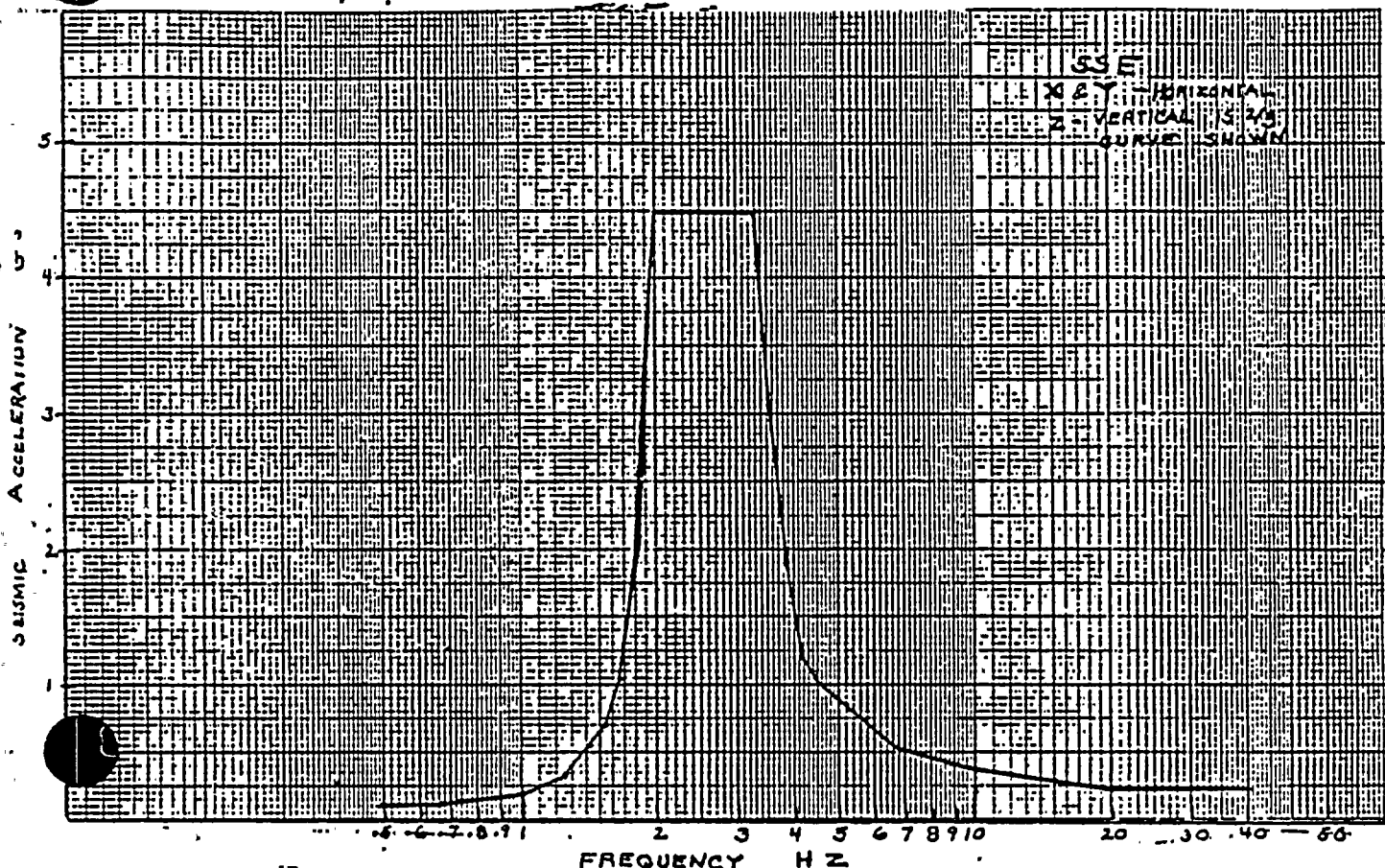


TABLE A2  
SSE

POINT	FREQ HZ	X OR Y		Z		POINT	FREQ HZ	X OR Y		Z	
		G's	in/sec <sup>2</sup>	G's	in/sec <sup>2</sup>			G's	in/sec <sup>2</sup>	G's	in/sec <sup>2</sup>
1	0.001	.001	.4	.001	.4	11	3.57	2.9	1121.	1.9	734.2
2	.50	.11	42.5	.07	27.0	12	3.85	1.9	734.2	1.3	502.3
3	.67	.12	46.4	.08	30.9	13	4.17	1.2	463.7	.80	309.1
4	1.0	.21	81.1	.14	54.1	14	4.45	1.05	405.7	.70	270.5
5	1.25	.34	131.4	.23	88.9	15	5.0	.83	320.7	.55	212.5
6	1.42	.54	208.7	.36	139.1	16	6.67	.54	208.7	.36	139.1
7	1.67	1.05	405.7	.70	270.5	17	10.	.36	139.1	.24	92.7
8	1.81	2.0	772.8	1.3	502.3	18	20.	.23	88.9	.15	58.0
9	2.0	4.5	1739.	3.0	1159.	19	33.	.23	88.9	.15	58.0
10	3.33	4.5	1739.	3.0	1159.	20	50.	.23	88.9	.15	58.0

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TABLE A3

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAOMMUO

MODE	FREQUENCY HZ	MODE COEFFICIENT FOR SPECIFIED DIRECTION		
		X	Y	Z
1	2.02	0.0512 *	128.5000 * MAX	0.0921 *
2	4.13	0.9912 * MAX	0.5329 *	4.3220 *
3	4.16	0.1254 *	0.3077 *	6.1500 * MAX
4	7.97	0.0356 *	0.0873 *	0.0010
5	9.05	0.0202 *	0.0218	0.0022
6	9.61	0.0235 *	0.0004	0.0015
7	11.58	0.0138 *	0.0239	0.0015
8	13.30	0.0229 *	0.0069	0.0003
9	14.91	0.1403 *	0.0002	0.0007
10	15.76	0.0560 *	0.0011	0.0111 *
11	18.05	0.0034	0.0097	0.0015
12	23.32	0.0006	0.0057	0.0003
13	28.73	0.0002	0.0006	0.0000
14	31.28	0.0040	0.0002	0.0008
15	35.25	0.0007	0.0002	0.0001
16	43.69	0.0001	0.0000	0.0001
17	54.53	0.0003	0.0001	0.0000
18	56.18	0.0003	0.0000	0.0001
19	61.02	0.0003	0.0000	0.0002
20	68.95	0.0003	0.0000	0.0004
21	75.35	0.0001	0.0000	0.0001
22	83.28	0.0001	0.0001	0.0001
23	85.98	0.0001	0.0000	0.0001
24	90.24	0.0001	0.0000	0.0001
25	91.85	0.0000	0.0000	0.0001
26	91.91	0.0000	0.0000	0.0002
27	123.90	0.0000	0.0000	0.0000
28	174.80	0.0000	0.0000	0.0000
SIGNIFICANCE FACTOR		0.50%	0.05%	0.05%
* INDICATES EXPANDED MODE				



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TABLE A4

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAOMMUS

MODE	FREQUENCY HZ	MODE COEFFICIENT FOR SPECIFIED DIRECTION		
		X	Y	Z
1	2.02	0.0960 *	241.0000 * MAX	0.1727 *
2	4.13	1.7650 * MAX	0.9486 *	7.7640 *
3	4.16	0.2216 *	0.5438 *	10.9500 * MAX
4	7.97	0.0817 *	0.2007 *	0.0022
5	9.05	0.0445 *	0.0477	0.0049
6	9.61	0.0504 *	0.0009	0.0032
7	11.58	0.0281 *	0.0487	0.0032
8	13.30	0.0450 *	0.0135	0.0007
9	14.91	0.2681 *	0.0004	0.0014
10	15.76	0.1054 *	0.0021	0.0204 *
11	18.05	0.0062	0.0176	0.0027
12	23.32	0.0012	0.0104	0.0005
13	28.73	0.0004	0.0011	0.0000
14	31.28	0.0076	0.0003	0.0016
15	35.25	0.0013	0.0003	0.0003
16	43.57	0.0002	0.0001	0.0003
17	54.53	0.0005	0.0003	0.0001
18	56.18	0.0005	0.0001	0.0002
19	61.02	0.0005	0.0000	0.0004
20	68.95	0.0005	0.0001	0.0008
21	75.35	0.0002	0.0000	0.0002
22	83.28	0.0002	0.0001	0.0001
23	85.98	0.0002	0.0001	0.0002
24	90.24	0.0001	0.0000	0.0001
25	91.85	0.0000	0.0000	0.0001
26	91.91	0.0000	0.0000	0.0003
27	123.90	0.0000	0.0000	0.0000
28	174.80	0.0000	0.0000	0.0000
SIGNIFICANCE FACTOR		0.50%	0.05%	0.05%
* INDICATES EXPANDED MODE				

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TABLE A5

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAOMMO

MODE	FREQUENCY HZ	MODE COEFFICIENT FOR SPECIFIED DIRECTION			
		X	Y	Z	
1	2.02	0.0510 *	128.5000 * MAX	0.1024 *	
2	2.93	1.8050 * MAX	0.0815 *	43.3700 * MAX	
3	4.14	0.7464 *	0.6138 *	0.0209	
4	6.34	0.1275 *	0.0140	0.6413 *	
5	7.99	0.0450 *	0.0865 *	0.0264 *	
6	9.05	0.0201 *	0.0219	0.0024	
7	9.54	0.0175 *	0.0008	0.0102	
8	11.59	0.0150 *	0.0237	0.0003	
9	13.31	0.0205 *	0.0068	0.0012	
10	15.09	0.1495 *	0.0006	0.0070	
11	18.04	0.0030	0.0097	0.0012	
12	23.32	0.0006	0.0057	0.0003	
13	28.72	0.0002	0.0006	0.0000	
14	31.23	0.0040	0.0002	0.0008	
15	35.24	0.0006	0.0002	0.0001	
16	43.59	0.0001	0.0000	0.0001	
17	54.52	0.0003	0.0001	0.0000	
18	56.18	0.0003	0.0000	0.0001	
19	61.02	0.0003	0.0000	0.0002	
20	68.89	0.0003	0.0000	0.0004	
21	75.33	0.0001	0.0000	0.0001	
22	83.28	0.0001	0.0001	0.0001	
23	85.97	0.0001	0.0000	0.0001	
24	90.22	0.0001	0.0000	0.0001	
25	91.85	0.0000	0.0000	0.0001	
26	91.91	0.0000	0.0000	0.0002	
27	123.90	0.0000	0.0000	0.0000	
28	174.80	0.0000	0.0000	0.0000	
SIGNIFICANCE FACTOR		0.50%	0.05%	0.05%	
* INDICATES EXPANDED MODE					

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TABLE A6

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PA0MMDS

MODE	FREQUENCY HZ	MODE COEFFICIENT FOR SPECIFIED DIRECTION		
		X	Y	Z
1	2.02	0.0957 *	241.0000 # MAX	0.1921 #
2	2.93	3.3850 * MAX	0.1528 #	81.3000 * MAX
3	4.14	1.3260 #	1.0900 #	0.0375
4	6.34	0.3128 #	0.0344	1.5370 #
5	7.99	0.1034 #	0.1987 #	0.0607 #
6	9.05	0.0441 #	0.0480	0.0054
7	9.64	0.0378 #	0.0017	0.0225
8	11.59	0.0325 #	0.0482	0.0007
9	13.31	0.0403 #	0.0135	0.0024
10	15.09	0.2849 #	0.0011	0.0129
11	18.04	0.0054	0.0177	0.0020
12	23.32	0.0012	0.0104	0.0006
13	28.72	0.0005	0.0011	0.0000
14	31.23	0.0077	0.0003	0.0015
15	35.24	0.0012	0.0003	0.0002
16	43.69	0.0002	0.0001	0.0003
17	54.52	0.0005	0.0003	0.0001
18	56.18	0.0005	0.0001	0.0002
19	61.02	0.0005	0.0000	0.0004
20	68.89	0.0005	0.0001	0.0008
21	75.33	0.0002	0.0000	0.0002
22	83.28	0.0002	0.0001	0.0001
23	85.97	0.0002	0.0001	0.0002
24	90.22	0.0001	0.0000	0.0001
25	91.85	0.0000	0.0000	0.0001
26	91.91	0.0000	0.0000	0.0003
27	123.90	0.0000	0.0000	0.0000
28	174.80	0.0000	0.0000	0.0000
SIGNIFICANCE FACTOR		0.50%	0.05%	0.05%
* INDICATES EXPANDED MODE				

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TABLE A7

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAONQUG  
 MODE FREQUENCY HZ MODE COEFFICIENT FOR SPECIFIED DIRECTION  
 X Y Z

MODE	FREQUENCY HZ	X	Y	Z
1	2.64	1.8160 * MAX	73.3000 * MAX	0.0730 *
2	4.16	0.5832 *	2.1760 *	0.0936 *
3	5.17	0.6651 *	0.0141	2.2470 * MAX
4	5.60	0.0695 *	0.2526 *	0.0327 *
5	7.44	0.0427 *	0.0315	0.0168 *
6	12.09	0.0184 *	0.0091	0.0013
7	14.04	0.0830 *	0.0039	0.0063 *
8	14.42	0.0446 *	0.0101	0.0038 *
9	16.64	0.0376 *	0.0031	0.0056 *
10	18.31	0.0654 *	0.0052	0.0157 *
11	18.63	0.0330 *	0.0084	0.0086 *
12	23.97	0.0004	0.0008	0.0012
13	24.79	0.0013	0.0036	0.0006
14	26.44	0.0028	0.0006	0.0023 *
15	37.79	0.0003	0.0001	0.0001
16	39.04	0.0002	0.0000	0.0004
17	45.33	0.0002	0.0001	0.0007
18	48.91	0.0001	0.0001	0.0007
19	53.25	0.0005	0.0001	0.0003
20	67.15	0.0002	0.0000	0.0000
21	68.43	0.0001	0.0000	0.0001
22	85.76	0.0000	0.0001	0.0001
23	87.93	0.0000	0.0000	0.0001
24	91.79	0.0000	0.0000	0.0000
25	91.94	0.0000	0.0000	0.0002
26	100.40	0.0000	0.0000	0.0001
27	121.00	0.0000	0.0000	0.0000
28	179.90	0.0000	0.0000	0.0000
SIGNIFICANCE FACTOR		0.50%	0.10%	0.10%
* INDICATES EXPANDED MODE				



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*CPH 8-25-87*

TABLE A 8

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAOMQUS

MODE	FREQUENCY HZ	MODE COEFFICIENT FOR SPECIFIED DIRECTION		
		X	Y	Z
1	2.64	3.4040 * MAX	137.4000 * MAX	0.1368 *
2	4.16	1.0310 *	3.8460 *	0.1667 *
3	5.17	1.6240 *	0.0345	5.3740 * MAX
4	5.60	0.1701 *	0.6180 *	0.0783 *
5	7.44	0.1006 *	0.0743	0.0393 *
6	12.09	0.0371 *	0.0183	0.0027
7	14.04	0.1611 *	0.0076	0.0120 *
8	14.42	0.0859 *	0.0194	0.0071 *
9	16.64	0.0698 *	0.0058	0.0101 *
10	18.31	0.1184 *	0.0095	0.0271 *
11	18.63	0.0594 *	0.0152	0.0147 *
12	23.97	0.0008	0.0015	0.0022
13	24.79	0.0025	0.0066	0.0010
14	26.44	0.0051	0.0011	0.0041
15	37.79	0.0006	0.0002	0.0001
16	39.04	0.0003	0.0001	0.0008
17	45.33	0.0003	0.0003	0.0014
18	48.91	0.0002	0.0002	0.0012
19	53.25	0.0010	0.0003	0.0005
20	67.15	0.0004	0.0000	0.0001
21	68.43	0.0001	0.0000	0.0003
22	85.76	0.0001	0.0001	0.0002
23	87.93	0.0000	0.0000	0.0003
24	91.79	0.0000	0.0000	0.0001
25	91.94	0.0000	0.0000	0.0003
26	100.40	0.0000	0.0000	0.0001
27	121.00	0.0000	0.0000	0.0000
28	179.90	0.0000	0.0000	0.0000

SIGNIFICANCE FACTOR 0.50%

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\* INDICATES EXPANDED MODE

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TABLE A 9

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAOMQDO  
 MODE FREQUENCY HZ MODE COEFFICIENT FOR SPECIFIED DIRECTION  
 X Y Z

MODE	FREQUENCY HZ	MODE COEFFICIENT X	MODE COEFFICIENT Y	MODE COEFFICIENT Z
1	2.64	1.8170 *	73.3000 * MAX	0.0943 *
2	3.14	3.4510 * MAX	0.0868 *	34.4800 * MAX
3	4.16	0.5675 *	2.1760 *	0.0089
4	5.60	0.0488 *	0.2528 *	0.0248
5	7.13	0.3278 *	0.0104	0.4583 *
6	7.47	0.0475 *	0.0298	0.1285 *
7	12.11	0.0300 *	0.0091	0.0001
8	14.37	0.0108	0.0107	0.0004
9	16.62	0.0257 *	0.0033	0.0032
10	17.54	0.0968 *	0.0021	0.0221
11	18.59	0.0120	0.0097	0.0034
12	23.88	0.0018	0.0006	0.0013
13	24.73	0.0022	0.0036	0.0006
14	26.04	0.0035	0.0009	0.0020
15	37.78	0.0003	0.0001	0.0001
16	39.03	0.0002	0.0000	0.0004
17	45.33	0.0002	0.0001	0.0008
18	48.89	0.0001	0.0001	0.0007
19	53.24	0.0005	0.0001	0.0003
20	67.15	0.0002	0.0000	0.0000
21	68.43	0.0001	0.0000	0.0001
22	85.76	0.0000	0.0001	0.0001
23	87.92	0.0000	0.0000	0.0001
24	91.79	0.0000	0.0000	0.0000
25	91.94	0.0000	0.0000	0.0002
26	100.40	0.0000	0.0000	0.0001
27	121.00	0.0000	0.0000	0.0000
28	179.90	0.0000	0.0000	0.0000
SIGNIFICANCE FACTOR		0.50%	0.10%	0.10%
* INDICATES EXPANDED MODE				

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TABLE A10

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAQMDS  
 MODE FREQUENCY HZ MODE COEFFICIENT FOR SPECIFIED DIRECTION  
 X Y Z

MODE	FREQUENCY HZ	MODE COEFFICIENT X	MODE COEFFICIENT Y	MODE COEFFICIENT Z
1	2.64	3.4070 *	137.4000 * MAX	0.1768 *
2	3.14	6.4710 * MAX	0.1628 *	64.6500 * MAX
3	4.16	1.0030 *	3.8460 *	0.0158
4	5.60	0.1195 *	0.6185 *	0.0594
5	7.13	0.7857 *	0.0249	1.0820 *
6	7.47	0.1118 *	0.0703	0.3002 *
7	12.11	0.0604 *	0.0183	0.0003
8	14.37	0.0209	0.0207	0.0007
9	16.62	0.0477 *	0.0061	0.0057
10	17.54	0.1774 *	0.0039	0.0387
11	18.59	0.0217	0.0174	0.0059
12	23.88	0.0033	0.0010	0.0022
13	24.73	0.0040	0.0065	0.0010
14	26.04	0.0064	0.0017	0.0035
15	37.78	0.0005	0.0002	0.0002
16	39.03	0.0003	0.0001	0.0008
17	45.33	0.0003	0.0003	0.0014
18	48.89	0.0002	0.0002	0.0012
19	53.24	0.0010	0.0003	0.0005
20	67.15	0.0004	0.0000	0.0001
21	68.43	0.0001	0.0000	0.0003
22	85.76	0.0001	0.0001	0.0002
23	87.92	0.0000	0.0000	0.0003
24	91.79	0.0000	0.0000	0.0001
25	91.94	0.0000	0.0000	0.0003
26	100.40	0.0000	0.0000	0.0001
27	121.00	0.0000	0.0000	0.0000
28	179.90	0.0000	0.0000	0.0000

SIGNIFICANCE FACTOR 0.50%

0.10%

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\* INDICATES EXPANDED MODE



*RAW 8.25.87*

TABLE A11

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PACMEUO  
 MODE FREQUENCY MODE COEFFICIENT FOR SPECIFIED DIRECTION  
 HZ X Y Z

1	3.49	1.7270 * MAX	22.4000 * MAX	0.2134 *
2	4.01	0.1411 *	5.5910 *	0.0283 *
3	4.75	0.1598 *	0.6807 *	0.0130 *
4	6.59	0.4757 *	0.0219	0.8717 * MAX
5	8.67	0.0222 *	0.2553 *	0.0176 *
6	11.56	0.0171 *	0.0277	0.0013
7	12.94	0.0715 *	0.0024	0.0310 *
8	13.40	0.0358 *	0.0200	0.0084 *
9	16.42	0.0265 *	0.0144	0.0051 *
10	18.18	0.0532 *	0.0048	0.0116 *
11	19.63	0.0502 *	0.0008	0.0133 *
12	23.63	0.0031	0.0005	0.0018 *
13	25.14	0.0010	0.0024	0.0005
14	31.32	0.0054	0.0000	0.0058 *
15	36.92	0.0003	0.0002	0.0016
16	42.67	0.0001	0.0001	0.0017
17	44.51	0.0001	0.0001	0.0026 *
18	49.81	0.0000	0.0000	0.0002
19	55.56	0.0004	0.0002	0.0004
20	66.86	0.0000	0.0001	0.0005
21	69.51	0.0001	0.0000	0.0000
22	78.16	0.0000	0.0000	0.0003
23	90.05	0.0000	0.0000	0.0000
24	91.70	0.0000	0.0000	0.0000
25	91.80	0.0000	0.0000	0.0002
26	92.80	0.0000	0.0000	0.0000
27	120.30	0.0000	0.0000	0.0000
28	128.20	0.0000	0.0000	0.0000
SIGNIFICANCE FACTOR		0.50%	0.20%	0.20%

\* INDICATES EXPANDED MODE.



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*ADD 8.25.87*

TABLE A12

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PADMEUS  
 MODE FREQUENCY HZ MODE COEFFICIENT FOR SPECIFIED DIRECTION  
 X Y Z

1	3.49	3.3090 * MAX	42.9300 * MAX	0.4038 *
2	4.01	0.2580 *	10.2300 *	0.0524 *
3	4.75	0.3754 *	1.5990 *	0.0303 *
4	6.59	1.1680 *	0.0538	2.0900 * MAX
5	8.67	0.0496 *	0.5695 *	0.0376 *
6	11.56	0.0348 *	0.0565	0.0026
7	12.94	0.1417 *	0.0048	0.0612 *
8	13.40	0.0704 *	0.0392	0.0164 *
9	16.42	0.0494 *	0.0268	0.0092 *
10	18.18	0.0765 *	0.0087	0.0201 *
11	19.63	0.0893 *	0.0014	0.0223 *
12	23.63	0.0056	0.0010	0.0031
13	25.14	0.0018	0.0045	0.0008
14	31.32	0.0103	0.0001	0.0108 *
15	36.92	0.0006	0.0003	0.0029
16	42.67	0.0002	0.0002	0.0032
17	44.51	0.0002	0.0002	0.0049 *
18	49.81	0.0000	0.0001	0.0004
19	55.56	0.0009	0.0004	0.0008
20	66.86	0.0001	0.0001	0.0010
21	69.51	0.0002	0.0000	0.0001
22	78.16	0.0000	0.0001	0.0006
23	90.05	0.0000	0.0000	0.0001
24	91.70	0.0000	0.0000	0.0000
25	91.80	0.0000	0.0000	0.0004
26	92.80	0.0001	0.0000	0.0001
27	120.30	0.0000	0.0000	0.0001
28	128.20	0.0001	0.0000	0.0001

SIGNIFICANCE FACTOR 0.50% 0.20% 0.20%

\* INDICATES EXPANDED MODE

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*0018.25.87*

TABLE A13

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAONEDO  
 MODE FREQUENCY HZ MODE COEFFICIENT FOR SPECIFIED DIRECTION  
 X Y Z

1	3.31	2.5940 * MAX	0.2644 *	28.0500 * MAX
2	3.49	1.7060 *	22.4000 * MAX	0.0088
3	4.01	0.1397 *	5.5910 *	0.0183
4	4.75	0.1593 *	0.6806 *	0.0155
5	7.82	0.3775 *	0.0425	0.3388 *
6	8.69	0.0100	0.2526 *	0.0332
7	11.56	0.0250 *	0.0274	0.0006
8	13.38	0.0204 *	0.0199	0.0011
9	16.38	0.0218 *	0.0146	0.0025
10	18.18	0.0539 *	0.0048	0.0120
11	19.54	0.0531 *	0.0009	0.0155
12	23.62	0.0033	0.0006	0.0022
13	25.14	0.0012	0.0024	0.0008
14	30.09	0.0063	0.0001	0.0074
15	36.88	0.0002	0.0002	0.0015
16	42.66	0.0001	0.0001	0.0017
17	44.44	0.0001	0.0001	0.0026
18	49.80	0.0000	0.0000	0.0002
19	55.56	0.0005	0.0002	0.0004
20	66.83	0.0000	0.0001	0.0005
21	69.51	0.0001	0.0000	0.0000
22	73.14	0.0000	0.0000	0.0003
23	90.05	0.0000	0.0000	0.0000
24	91.70	0.0000	0.0000	0.0000
25	91.80	0.0000	0.0000	0.0002
26	92.80	0.0000	0.0000	0.0000
27	120.30	0.0000	0.0000	0.0000
28	123.20	0.0000	0.0000	0.0000
SIGNIFICANCE FACTOR		0.50%	0.20%	0.20%

\* INDICATES EXPANDED MODE

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*RMA 9-25-87*

TABLE A 14

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAOMEDS

MODE FREQUENCY MODE COEFFICIENT FOR SPECIFIED DIRECTION

	HZ	X	Y	Z
1	3.31	4.8640 * MAX	0.4958 *	52.6000 * MAX
2	3.49	3.2680 *	42.9200 * MAX	0.0166
3	4.01	0.2555 *	10.2300 *	0.0340
4	4.75	0.3741 *	1.5990 *	0.0361
5	7.82	0.8746 *	0.0984 *	0.7828 *
6	8.69	0.0222	0.5631 *	0.0747
7	11.56	0.0510 *	0.0558	0.0012
8	13.38	0.0400 *	0.0391	0.0022
9	16.38	0.0407 *	0.0272	0.0044
10	18.18	0.0979 *	0.0086	0.0207
11	19.54	0.0946 *	0.0017	0.0260
12	23.62	0.0060	0.0010	0.0038
13	25.14	0.0021	0.0045	0.0014
14	30.09	0.0120	0.0001	0.0135
15	36.88	0.0005	0.0003	0.0028
16	42.66	0.0001	0.0002	0.0033
17	44.44	0.0001	0.0002	0.0049
18	49.80	0.0000	0.0001	0.0004
19	55.56	0.0009	0.0004	0.0008
20	66.83	0.0001	0.0001	0.0010
21	69.51	0.0002	0.0000	0.0001
22	78.14	0.0000	0.0001	0.0006
23	90.05	0.0000	0.0000	0.0001
24	91.70	0.0000	0.0000	0.0000
25	91.80	0.0000	0.0000	0.0004
26	92.80	0.0001	0.0000	0.0001
27	120.30	0.0000	0.0000	0.0001
28	128.20	0.0001	0.0000	0.0001

SIGNIFICANCE FACTOR 0.50%

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0.20%

\* INDICATES EXPANDED MODE

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TABLE A 15

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAOMNO  
 MODE FREQUENCY HZ MODE COEFFICIENT FOR SPECIFIED DIRECTION  
 X Y Z

1	2.02	0.0515 *	128.5000 * MAX	0.0772 *
2	4.14	0.7560 * MAX	0.6143 *	0.1332 *
3	5.19	0.2460 *	0.0105	1.9000 * MAX
4	7.98	0.0403 *	0.0870 *	0.0135 *
5	9.05	0.0201 *	0.0219	0.0023 *
6	9.63	0.0197 *	0.0004	0.0072 *
7	11.59	0.0155 *	0.0237	0.0001
8	13.31	0.0208 *	0.0068	0.0011
9	15.09	0.1499 *	0.0006	0.0067 *
10	18.04	0.0029	0.0098	0.0011
11	23.32	0.0006	0.0057	0.0003
12	28.66	0.0005	0.0006	0.0001
13	29.99	0.0023	0.0002	0.0000
14	31.77	0.0033	0.0001	0.0009
15	35.30	0.0007	0.0001	0.0002
16	43.69	0.0001	0.0000	0.0001
17	54.53	0.0003	0.0001	0.0000
18	56.18	0.0003	0.0000	0.0001
19	61.02	0.0003	0.0000	0.0002
20	68.96	0.0003	0.0000	0.0004
21	75.35	0.0001	0.0000	0.0001
22	83.28	0.0001	0.0001	0.0001
23	85.99	0.0001	0.0000	0.0001
24	90.24	0.0001	0.0000	0.0001
25	91.85	0.0000	0.0000	0.0001
26	91.91	0.0000	0.0000	0.0002
27	123.90	0.0000	0.0000	0.0000
28	174.80	0.0000	0.0000	0.0000
SIGNIFICANCE FACTOR		0.50%	0.05%	0.10%
* INDICATES EXPANDED MODE				



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TABLE A16

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAOMMNS  
 MODE FREQUENCY MODE COEFFICIENT FOR SPECIFIED DIRECTION  
 HZ X Y Z

1	2.02	0.0966 *	241.0000 * MAX	0.1447 *
2	4.14	1.3430 * MAX	1.0910 *	0.2387 *
3	5.19	0.6009 *	0.0256	4.5430 * MAX
4	7.98	0.0926 *	0.2001 *	0.0311 *
5	9.05	0.0442 *	0.0480	0.0052 *
6	9.63	0.0422 *	0.0008	0.0159 *
7	11.59	0.0316 *	0.0483	0.0002
8	13.31	0.0409 *	0.0135	0.0022
9	15.09	0.2855 *	0.0011	0.0125 *
10	18.04	0.0052	0.0177	0.0019
11	23.32	0.0012	0.0104	0.0006
12	28.66	0.0010	0.0011	0.0002
13	29.99	0.0044	0.0003	0.0001
14	31.77	0.0062	0.0002	0.0017
15	35.30	0.0014	0.0003	0.0003
16	43.69	0.0002	0.0001	0.0003
17	54.53	0.0005	0.0003	0.0001
18	56.18	0.0005	0.0001	0.0002
19	61.02	0.0006	0.0000	0.0004
20	68.96	0.0005	0.0001	0.0008
21	75.35	0.0002	0.0000	0.0002
22	83.28	0.0002	0.0001	0.0001
23	85.99	0.0002	0.0001	0.0002
24	90.24	0.0001	0.0000	0.0001
25	91.85	0.0000	0.0000	0.0001
26	91.91	0.0000	0.0000	0.0003
27	123.90	0.0000	0.0000	0.0000
28	174.80	0.0000	0.0000	0.0000
SIGNIFICANCE FACTOR		0.50%	0.05%	0.10%
* INDICATES EXPANDED MODE				

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TABLE A17

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAOMQNO  
 MODE FREQUENCY MODE COEFFICIENT FOR SPECIFIED DIRECTION  
 HZ X Y Z

1	2.64	1.8150 * MAX	73.3000 * MAX	0.0630 *
2	4.16	0.5746 *	2.1760 *	0.0471 *
3	5.60	0.0414 *	0.2528 *	0.0453 *
4	6.33	0.4446 *	0.0052	0.9009 * MAX
5	7.44	0.0220 *	0.0314	0.0432 *
6	12.11	0.0284 *	0.0091	0.0003
7	14.36	0.0098 *	0.0107	0.0003
8	16.61	0.0206 *	0.0034	0.0022 *
9	17.34	0.1015 *	0.0018	0.0226 *
10	18.58	0.0100 *	0.0097	0.0030 *
11	23.73	0.0040	0.0002	0.0011
12	24.60	0.0035	0.0033	0.0004
13	25.53	0.0038	0.0017	0.0011
14	31.39	0.0002	0.0000	0.0011
15	37.79	0.0003	0.0001	0.0001
16	39.05	0.0001	0.0000	0.0004
17	45.34	0.0002	0.0001	0.0007
18	48.92	0.0001	0.0001	0.0006
19	53.26	0.0005	0.0001	0.0003
20	67.15	0.0002	0.0000	0.0000
21	68.43	0.0001	0.0000	0.0001
22	85.76	0.0000	0.0001	0.0001
23	87.93	0.0000	0.0000	0.0001
24	91.79	0.0000	0.0000	0.0000
25	91.94	0.0000	0.0000	0.0002
26	100.40	0.0000	0.0000	0.0001
27	121.00	0.0000	0.0000	0.0000
28	179.90	0.0000	0.0000	0.0000

SIGNIFICANCE FACTOR 0.50%

0.10%

0.20%

\* INDICATES EXPANDED MODE

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TABLE A 18

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAOMQNS					
MODE	FREQUENCY HZ	MODE COEFFICIENT FOR SPECIFIED DIRECTION			
		X	Y	Z	
1	2.64	3.4030 * MAX	137.4000 * MAX	0.1181 *	
2	4.16	1.0160 *	3.8460 *	0.0839 *	
3	5.60	0.1014 *	0.6185 *	0.1085 *	
4	6.33	1.0910 *	0.0127	2.1590 * MAX	
5	7.44	0.0520 *	0.0740	0.1010 *	
6	12.11	0.0572 *	0.0183	0.0006	
7	14.36	0.0189 *	0.0207	0.0005	
8	16.61	0.0382 *	0.0063	0.0039	
9	17.34	0.1864 *	0.0033	0.0398 *	
10	18.58	0.0181 *	0.0175	0.0051 *	
11	23.73	0.0072	0.0003	0.0020	
12	24.60	0.0063	0.0061	0.0007	
13	25.53	0.0070	0.0031	0.0020	
14	31.39	0.0004	0.0000	0.0021	
15	37.79	0.0006	0.0002	0.0001	
16	39.05	0.0003	0.0001	0.0008	
17	45.34	0.0003	0.0003	0.0014	
18	48.92	0.0002	0.0002	0.0012	
19	53.26	0.0010	0.0003	0.0005	
20	67.15	0.0004	0.0000	0.0001	
21	68.43	0.0001	0.0000	0.0003	
22	85.76	0.0001	0.0001	0.0002	
23	87.93	0.0000	0.0000	0.0003	
24	91.79	0.0000	0.0000	0.0001	
25	91.94	0.0000	0.0000	0.0003	
26	100.40	0.0000	0.0000	0.0001	
27	121.00	0.0000	0.0000	0.0000	
28	179.90	0.0000	0.0000	0.0000	
SIGNIFICANCE FACTOR		0.50%	0.10%	0.20%	
* INDICATES EXPANDED MODE					

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TABLE A19

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAOMRNO

MODE	FREQUENCY HZ	MODE COEFFICIENT FOR SPECIFIED DIRECTION		
		X	Y	Z
1	3.61	1.2600 * MAX	9.8790 * MAX	0.1846 *
2	4.06	0.2330 *	6.7300 *	0.0235 *
3	4.65	0.2234 *	0.5985 *	0.0079 *
4	7.74	0.4103 *	0.0177	0.3795 * MAX
5	10.73	0.0098 *	0.1372 *	0.0049 *
6	11.45	0.0223 *	0.1251 *	0.0035 *
7	13.20	0.0208 *	0.0255 *	0.0015
8	15.91	0.0269 *	0.0182	0.0026 *
9	17.76	0.0426 *	0.0072	0.0089 *
10	19.31	0.0602 *	0.0003	0.0171 *
11	22.81	0.0040	0.0003	0.0025 *
12	24.67	0.0012	0.0020	0.0008
13	27.08	0.0062	0.0000	0.0088 *
14	34.26	0.0029	0.0002	0.0008
15	36.49	0.0022	0.0003	0.0015 *
16	42.78	0.0005	0.0001	0.0014
17	47.13	0.0008	0.0001	0.0036 *
18	52.26	0.0002	0.0001	0.0005
19	54.74	0.0002	0.0002	0.0011
20	63.78	0.0002	0.0001	0.0008
21	69.25	0.0001	0.0000	0.0004
22	73.82	0.0000	0.0000	0.0004
23	86.27	0.0000	0.0000	0.0001
24	91.78	0.0000	0.0000	0.0001
25	92.03	0.0000	0.0000	0.0001
26	96.97	0.0000	0.0000	0.0001
27	116.40	0.0000	0.0000	0.0001
28	124.10	0.0000	0.0000	0.0000

SIGNIFICANCE FACTOR 0.50%      0.20%      0.40%

\* INDICATES EXPANDED MODE



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*OWD 8.25.87*

TABLE A 20

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PADMRNS  
 MODE FREQUENCY HZ MODE COEFFICIENT FOR SPECIFIED DIRECTION  
 X Y Z

1	3.61	2.4290 * MAX	19.0500 * MAX	0.3518 *
2	4.06	0.4212 *	12.1600 *	0.0429 *
3	4.65	0.5161 *	1.3830 *	0.0183 *
4	7.74	0.9540 *	0.0411 *	0.8788 * MAX
5	10.73	0.0203 *	0.2852 *	0.0104 *
6	11.45	0.0455 *	0.2558 *	0.0073 *
7	13.20	0.0410 *	0.0503 *	0.0029
8	15.91	0.0506 *	0.0341	0.0048 *
9	17.76	0.0779 *	0.0132	0.0156 *
10	19.31	0.1076 *	0.0005	0.0290 *
11	22.81	0.0073	0.0005	0.0042 *
12	24.67	0.0022	0.0037	0.0014
13	27.08	0.0113	0.0001	0.0157 *
14	34.26	0.0057	0.0004	0.0014
15	36.49	0.0042	0.0006	0.0029
16	42.78	0.0010	0.0003	0.0027
17	47.13	0.0015	0.0002	0.0067 *
18	52.26	0.0003	0.0001	0.0009
19	54.74	0.0004	0.0004	0.0021
20	63.78	0.0003	0.0001	0.0016
21	69.25	0.0002	0.0001	0.0008
22	73.82	0.0001	0.0000	0.0008
23	86.27	0.0001	0.0001	0.0002
24	91.78	0.0000	0.0000	0.0003
25	92.03	0.0000	0.0000	0.0002
26	96.97	0.0001	0.0000	0.0002
27	116.40	0.0001	0.0000	0.0002
28	124.10	0.0000	0.0000	0.0001
SIGNIFICANCE FACTOR		0.50%	0.20%	0.40%
* INDICATES EXPANDED MODE				

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WHITING REQ. 79604 DATE 6-17-87  
 BY MJM PAGE A-22 OF 22  
*JUN 23 87*

TABLE A 21

SUMMARY OF COMPUTER RUNS

OBE

TROLLEY	LOAD	STATIC	DYNAMIC	SUM	
				STRAIGHT	FACTORED
MID	50T UP	AOMMU	AOMMUO	PSAOMMUO	PFAOMMUO
	50T DN	AOMMU	AOMMDO	PSAOMMDO	PFAOMMDO
	NO LOAD	AOMMN	AOMMNO	PSAOMMNO	PFAOMMNO
1/4	50T UP	AOMQU	AOMQUO	PSAOMQUO	PFAOMQUO
	50T DN	AOMQU	AOMQDO	PSAOMQDO	PFAOMQDO
	NO LOAD	AOMQN	AOMQNO	PSAOMQNO	PFAOMQNO
END	50T UP	AOMEU	AOMEUO	PSAOMEUO	PFAOMEUO
	50T DN	AOMEU	AOMEDO	PSAOMEDO	PFAOMEDO
	NO LOAD	AOMRN	AOMRNO	PSAOMRNO	PFAOMRNS

SSE

TROLLEY	LOAD	STATIC	DYNAMIC	SUM	
				STRAIGHT	FACTORED
MID	50T UP	AOMMU	AOMMUS	PSAOMMUS	PFAOMMUS
	50T DN	AOMMU	AOMMDS	PSAOMMDS	PFAOMMDS
	NO LOAD	AOMMN	AOMMNS	PSAOMMNS	PFAOMMNS
1/4	50T UP	AOMQU	AOMQUS	PSAOMQUS	PFAOMQUS
	50T DN	AOMQU	AOMQDS	PSAOMQDS	PFAOMQDS
	NO LOAD	AOMQN	AOMQNS	PSAOMQNS	PFAOMQNS
END	50T UP	AOMEU	AOMEUS	PSAOMEUS	PFAOMEUS
	50T DN	AOMEU	AOMEDS	PSAOMEDS	PFAOMEDS
	NO LOAD	AOMRN	AOMRNS	PSAOMRNS	PFAOMRNS

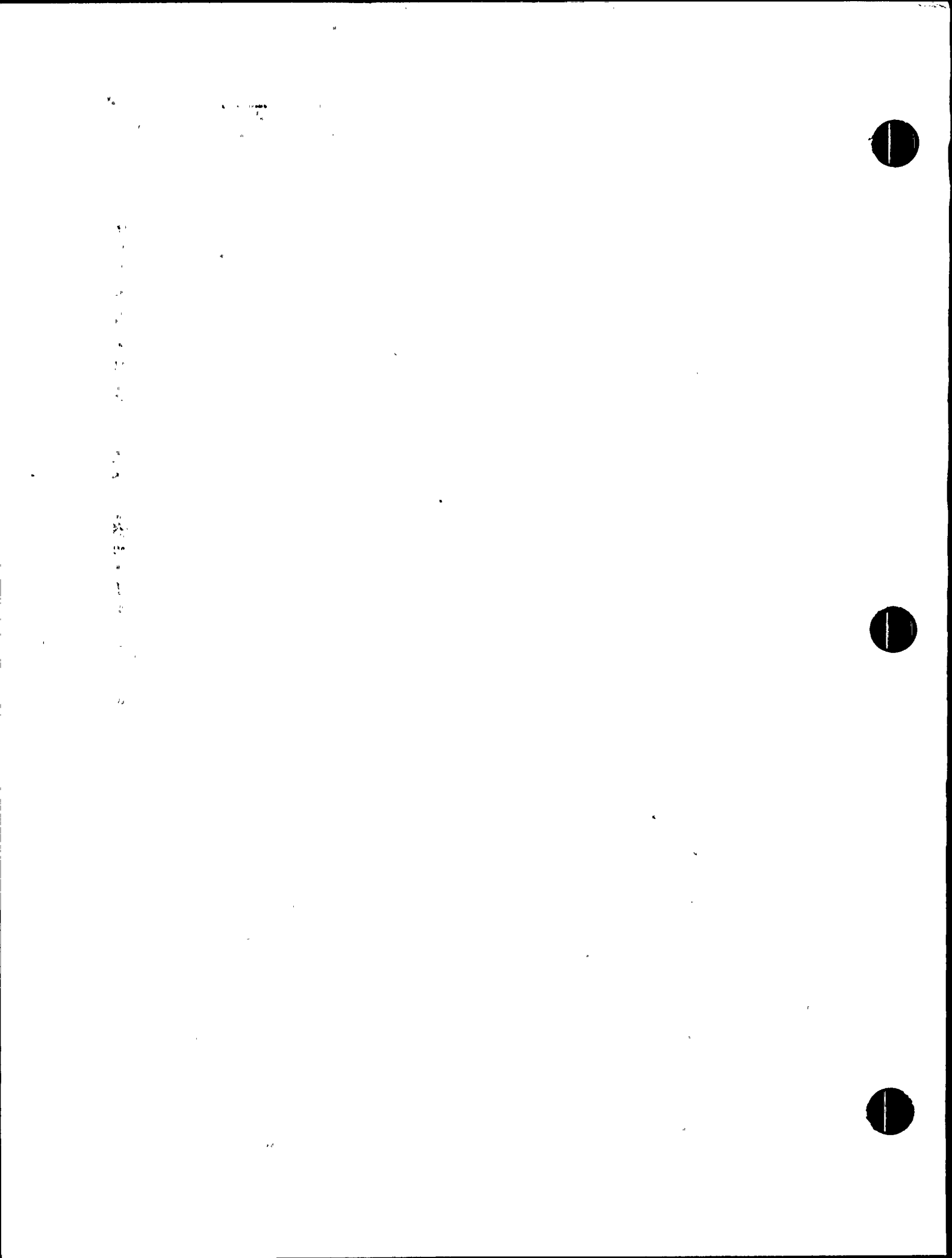
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WHITING REQN. 79604 DATE 8-21-87  
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APPENDIX B

This appendix summarizes the maximum stresses and maximum loadings from the computer output. All values are after application of scale factor for slip as explained in section 4 except for tables B19 to B54 which show factored and unfactored reactions.

PAGE	TABLE	TITLE
B-3	B1	Max Stresses, Mid, 50T UP, OBE
	B2	Max Stresses, Mid, 50T DN, OBE
	B3	Max Stresses, Mid, No LD, OBE
B-4	B4	Max Stresses, Mid, 50T UP, SSE
	B5	Max Stresses, Mid, 50T DN, SSE
	B6	Max Stresses, Mid, NO LD, SSE
B-5	B7	Max Stresses, 1/4, 50T UP, OBE
	B8	Max Stresses, 1/4, 50T DN, OBE
	B9	Max Stresses, 1/4, NO LD, OBE
B-6	B10	Max Stresses, 1/4, 50T UP, SSE
	B11	Max Stresses, 1/4, 50T DN, SSE
	B12	Max Stresses, 1/4, NO LD, SSE
B-7	B13	Max Stresses, END, 50T UP, OBE
	B14	Max Stresses, END, 50T DN, OBE
	B15	Max Stresses, RHE, NO LOAD, OBE
B-8	B16	Max Stresses, END, 50T UP, SSE
	B17	Max Stresses, END, 50T DN, SSE
	B18	Max Stresses, RHE, NO LD, SSE
B-9	B19	Reactions, Mid, 50T UP, OBE
B-10	B20	Reactions, Mid, 50T UP, OBE, SCALED
B-11	B21	Reactions, Mid, 50T UP, SSE
B-12	B22	Reactions, Mid, 50T UP, SSE, SCALED
B-13	B23	Reactions, Mid, 50T DN, OBE
B-14	B24	Reactions, Mid, 50T DN, OBE, SCALED
B-15	B25	Reactions, Mid, 50T DN, SSE
B-16	B26	Reactions, Mid, 50T DN, SSE, SCALED
B-17	B27	Reactions, 1/4, 50T UP, OBE
B-18	B28	Reactions, 1/4, 50T UP, OBE, SCALED
B-19	B29	Reactions, 1/4, 50T UP, SSE
B-20	B30	Reactions, 1/4, 50T UP, SSE, SCALED
B-21	B31	Reactions, 1/4, 50T DN, OBE
B-22	B32	Reactions, 1/4, 50T DN, OBE, SCALED
B-23	B33	Reactions, 1/4, 50T DN, SSE
B-24	B34	Reactions, 1/4, 50T DN, SSE, SCALED
B-25	B35	Reactions, END, 50T UP, OBE
B-26	B36	Reactions, END, 50T UP, OBE, SCALED
B-27	B37	Reactions, END, 50T UP, SSE
B-28	B38	Reactions, END, 50T UP, SSE, SCALED
B-29	B39	Reactions, END, 50T DN, OBE
B-30	B40	Reactions, END, 50T DN, OBE, SCALED



WHITING REQ. 79604 DATE 8-21-87  
 BY MJM PAGE 82 OF 58  
*RA 8.24.87*

B-31	B41	Reactions, END, 50T DN, SSE
B-32	B42	Reactions, END, 50T DN, SSE, SCALED
B-33	B43	Reactions, MID, No Load, OBE
B-34	B44	Reactions, MID, No Load, OBE, SCALED
B-35	B45	Reactions, MID, No Load, SSE
B-36	B46	Reactions, MID, No Load, SSE, SCALED
B-37	B47	Reactions, 1/4, No Load, OBE
B-38	B48	Reactions, 1/4, No Load, OBE, SCALED
B-39	B49	Reactions, 1/4, No Load, SSE
B-40	B50	Reactions, 1/4, No Load, SSE, SCALED
B-41	B51	Reactions, END, No Load, OBE
B-42	B52	Reactions, END, No Load, OBE, SCALED
B-43	B53	Reactions, END, No Load, SSE
B-44	B54	Reactions, END, No Load, SSE, SCALED
B-45	B55	Girder to End Tie Connection, OBE
B-46	B56	Girder to End Tie Connection, SSE
B-47	B57	Trolley Reactions, OBE, Sum
B-48	B58	Trolley Reactions, OBE, Diff
B-49	B59	Trolley Reactions, SSE, Sum
B-50	B60	Trolley Reactions, SSE, Diff
B-51	B61	Rope Loads, OBE, Sum
B-52	B62	Rope Loads, OBE, Diff
B-53	B63	Rope Loads, SSE, Sum
B-54	B64	Rope Loads, SSE, Diff
B-55	B65	Element Load, Girder A at MAX STRESS
B-56	B66	Element Load, Girder B at MAX STRESS
B-57	B67	Element Load, Girder End, OBE
B-58	B68	Element Load, Girder End, SSE



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TABLE 01

MAXIMUM STRESSES FROM PEAOQMUQ  
 OBE MID 50 U

COMPONENT	ELEM	NODE	X	Y	Z	SRSS	STATIC	SUM
GIRDER A	28	311	793.	3120.	7080.	7778.	9977.	17755.
GIRDER B	52	361	463.	4813.	4471.	6586.	9304.	15890.
END CONNECT-RHE	17	154	1115.	9524.	8691.	12942.	505.	13447.
END CONNECT-LHE	74	252	1282.	8105.	9979.	12920.	306.	13225.

TABLE 02

MAXIMUM STRESSES FROM PEAOQMDQ  
 OBE MID 50 D

COMPONENT	ELEM	NODE	X	Y	Z	SRSS	STATIC	SUM
GIRDER A	28	312	567.	5190.	10826.	12019.	9827.	21846.
GIRDER B	52	361	530.	6222.	10468.	12189.	9304.	21493.
END CONNECT-RHE	17	154	932.	12324.	195.	12361.	505.	12866.
END CONNECT-LHE	74	252	1070.	10476.	269.	10534.	306.	10840.

TABLE 03

MAXIMUM STRESSES FROM PEAOQMNQ  
 OBE MID NL

COMPONENT	ELEM	NODE	X	Y	Z	SRSS	STATIC	SUM
GIRDER A	30	313	249.	3325.	1442.	3633.	6483.	10116.
GIRDER B	52	361	336.	3510.	1351.	3776.	6248.	10025.
END CONNECT-RHE	17	154	944.	6980.	194.	7046.	497.	7543.
END CONNECT-LHE	74	252	1084.	5953.	208.	6055.	315.	6370.

STRESS IN PSI

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TABLE 84

MAXIMUM STRESSES FROM PFAQMMUS  
 SSE MID 50 U

COMPONENT	ELEM	NODE	X	Y	Z	SRSS	STATIC	SUM
GIRDER A	28	311	1425.	4036.	12653.	13357.	9977.	23335.
GIRDER B	54	363	1049.	4178.	10421.	11276.	8853.	20129.
END CONNECT-RHE	17	154	1988.	12140.	15545.	19836.	505.	20341.
END CONNECT-LHE	74	252	2285.	10374.	17850.	20771.	306.	21077.

TABLE 85

MAXIMUM STRESSES FROM PFAQMMDS  
 SSE MID 50 D

COMPONENT	ELEM	NODE	X	Y	Z	SRSS	STATIC	SUM
GIRDER A	28	312	1074.	7391.	20314.	21643.	9827.	31470.
GIRDER B	52	361	1009.	8858.	19636.	21565.	9304.	30869.
END CONNECT-RHE	17	154	1660.	17562.	369.	17644.	505.	18149.
END CONNECT-LHE	74	252	1904.	14942.	506.	15071.	306.	15377.

TABLE 86

MAXIMUM STRESSES FROM PFAQMMNS  
 SSE MID NL

COMPONENT	ELEM	NODE	X	Y	Z	SRSS	STATIC	SUM
GIRDER A	28	312	781.	3811.	3625.	5318.	6769.	12087.
GIRDER B	52	361	695.	4555.	3230.	5627.	6248.	11876.
END CONNECT-RHE	17	154	1681.	9099.	377.	9261.	497.	9758.
END CONNECT-LHE	74	252	1929.	7794.	389.	8039.	315.	8354.

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TABLE 87

MAXIMUM STRESSES FROM PFAOMQUO

OBE 1/4 50 U

COMPONENT	ELEM	NODE	X	Y	Z	SRSS	STATIC	SUM
GIRDER A	27	310	901.	2878.	1665.	3445.	7184.	10630.
GIRDER B	51	360	806.	2481.	1633.	3078.	7267.	10344.
END CONNECT-RHE	17	154	1797.	7141.	201.	7367.	489.	7856.
END CONNECT-LHE	74	252	1769.	7128.	184.	7347.	351.	7698.

TABLE 88

MAXIMUM STRESSES FROM PFAOMQDO

OBE 1/4 50 D

COMPONENT	ELEM	NODE	X	Y	Z	SRSS	STATIC	SUM
GIRDER A	27	310	1053.	4014.	7040.	8172.	7184.	15356.
GIRDER B	51	360	989.	3478.	7073.	7943.	7267.	15210.
END CONNECT-RHE	17	154	1798.	9233.	632.	9427.	489.	9917.
END CONNECT-LHE	74	252	1760.	9800.	525.	9971.	351.	10322.

TABLE 89

MAXIMUM STRESSES FROM PFAOMQNO

OBE 1/4 NL

COMPONENT	ELEM	NODE	X	Y	Z	SRSS	STATIC	SUM
GIRDER A	26	310	848.	2199.	778.	2482.	4778.	7260.
GIRDER B	51	360	734.	1878.	818.	2176.	4860.	7036.
END CONNECT-RHE	17	154	1779.	5989.	194.	6250.	478.	6728.
END CONNECT-LHE	74	252	1739.	5552.	144.	5820.	357.	6176.

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TABLE 810

MAXIMUM STRESSES FROM PFAOMQUS  
 SSE 1/4 50 U

COMPONENT	ELEM NODE		X	Y	Z	SRSS	STATIC	SUM
GIRDER A	27	310	1895.	3678.	3982.	5742.	7184.	12927
GIRDER B	51	360	1695.	3120.	3905.	5278.	7267.	12544
END CONNECT-RHE	17	154	3310.	10232.	384.	10761.	489.	11250
END CONNECT-LHE	74	252	3308.	9315.	393.	9892.	351.	10244

TABLE 811

MAXIMUM STRESSES FROM PFAOMQDS  
 SSE 1/4 50 D

COMPONENT	ELEM NODE		X	Y	Z	SRSS	STATIC	SUM
GIRDER A	27	310	2035.	5665.	13219.	14525.	7184.	21709.
GIRDER B	51	360	1914.	4878.	13277.	14274.	7267.	21541.
END CONNECT-RHE	17	154	3342.	13631.	1433.	14108.	489.	14597.
END CONNECT-LHE	74	252	3311.	13926.	1222.	14366.	351.	14717.

TABLE 812

MAXIMUM STRESSES FROM PFAOMQNS  
 SSE 1/4 NL

COMPONENT	ELEM NODE		X	Y	Z	SRSS	STATIC	SUM
GIRDER A	27	310	1649.	2857.	1931.	3822.	4777.	8600.
GIRDER B	51	360	1515.	2371.	1959.	3429.	4860.	8289.
END CONNECT-RHE	17	154	3275.	9002.	434.	9589.	478.	10067.
END CONNECT-LHE	74	252	3248.	7451.	321.	8134.	357.	8491.

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*8-25-37*

TABLE B13

MAXIMUM STRESSES FROM PEAQMEUQ

OBE END 50 U

COMPONENT	ELEM NODE		X	Y	Z	SRSS	STATIC	SUM
GIRDER A	29	312	962.	2559.	691.	2820.	4142.	6962.
GIRDER B	43	353	799.	6909.	388.	6966.	2283.	9249.
END CONNECT-RHE	17	154	493.	10150.	156.	10163.	470.	10633.
END CONNECT-LHE	75	254	1297.	9558.	185.	9647.	445.	10092.

TABLE B14

MAXIMUM STRESSES FROM PEAQMEDD

OBE END 50 D

COMPONENT	ELEM NODE		X	Y	Z	SRSS	STATIC	SUM
GIRDER A	35	319	2181.	9504.	372.	9758.	178.	9936.
GIRDER B	43	353	940.	7280.	2032.	7616.	2283.	9899.
END CONNECT-RHE	17	154	498.	10203.	220.	10218.	470.	10687.
END CONNECT-LHE	74	252	2777.	12173.	384.	12491.	358.	12849.

TABLE B15

MAXIMUM STRESSES FROM PEAQMRND

OBE RHE NL

COMPONENT	ELEM NODE		X	Y	Z	SRSS	STATIC	SUM
GIRDER A	35	319	1516.	6905.	224.	7073.	142.	7215.
GIRDER B	59	369	583.	9791.	80.	9809.	155.	9963.
END CONNECT-RHE	17	154	1292.	10770.	192.	10849.	440.	11289.
END CONNECT-LHE	75	254	485.	12815.	56.	12824.	443.	13268.

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TABLE 016

MAXIMUM STRESSES FROM PFAOMEUS

COMPONENT	ELEM NODE		SSE END 50 U			SRSS	STATIC	SUM
	X	Y	Z					
GIRDER A	19	303	780.	6278.	998.	6405.	2508.	8913.
GIRDER B	42	351	815.	14159.	646.	14197.	529.	14727.
END CONNECT-RHE	17	154	975.	18612.	347.	18641.	470.	19111.
END CONNECT-LHE	75	254	2486.	16707.	357.	16895.	445.	17340.

TABLE 017

MAXIMUM STRESSES FROM PFAOMEDS

COMPONENT	ELEM NODE		SSE END 50 D			SRSS	STATIC	SUM
	X	Y	Z					
GIRDER A	35	319	4179.	12973.	699.	13647.	178.	13825.
GIRDER B	43	353	1808.	12783.	3815.	13462.	2283.	15745.
END CONNECT-RHE	17	154	1003.	18675.	472.	18708.	470.	19178.
END CONNECT-LHE	75	254	2454.	17781.	366.	17953.	445.	18398.

TABLE 018

MAXIMUM STRESSES FROM PFAOMRNS

COMPONENT	ELEM NODE		SSE RHE NL			SRSS	STATIC	SUM
	X	Y	Z					
GIRDER A	35	319	2922.	10877.	428.	11271.	142.	11413
GIRDER B	59	369	1102.	17599.	153.	17634.	155.	17789
END CONNECT-RHE	17	154	2505.	18890.	373.	19059.	440.	19499
END CONNECT-LHE	75	254	894.	23167.	104.	23185.	443.	23628

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1. The first part of the document is a list of names and titles.

The main body of the document contains several paragraphs of text, which are mostly illegible due to the low quality of the scan. The text appears to be a formal report or a list of items, but the specific details cannot be discerned.

79604/MJM/AEP, DCC, OLD MAIN TROLLEY MID, 50T LD UP / OBE 4W

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	707.	241646.	254.	241647.	0.	241647.	-241647.
101 FZ	4289.	31966.	39130.	50709.	102034.	152743.	51325.
101 MX	72713.	16139508.	441464.	16145709.	44267.	16189976.	-16101441.
102 FZ	3364.	31526.	35735.	47772.	93881.	141653.	46108.
102 MX	57126.	682900.	354863.	771715.	90260.	861974.	-681455.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	32513.	991.	8493.	33618.	0.	33618.	-33618.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	1088.	221007.	6955.	221119.	-0.	221119.	-221119.
201 FZ	3288.	27791.	37826.	47053.	98309.	145362.	51256.
201 MX	43592.	14919499.	129377.	14920124.	43387.	14963511.	-14876737.
202 FZ	3456.	27683.	34678.	44507.	90155.	134563.	45648.
202 MX	23884.	217245.	128539.	253551.	89376.	342927.	-164176.

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SRSS-4.3 WHITING CORPORATION ANSYS SRSS PROGRAM  
 TABLE # 020 LS 2 MODE 1 SCALE FACTOR = .1096

79604 87/06/12.  
 BY MJM PAGE 0-10 OF 58

79604/MJM/AEP, DCC, OLD MAIN TROLLEY MID, 50T LD UP / ORE 10

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	707.	26488.	254.	26498.	0.	26498.	-26498.
101 FZ	4289.	4613.	39130.	39634.	102034.	141668.	62400.
101 MX	72713.	1769535.	441464.	1825221.	44267.	1869489.	-1780954.
102 FZ	3384.	4272.	35735.	36146.	93881.	130027.	57734.
102 MX	57126.	81473.	354863.	368550.	90260.	458809.	-278290.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	32513.	973.	8493.	33518.	0.	33518.	-33518.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	1088.	24231.	6955.	25233.	-0.	25233.	-25233.
201 FZ	3288.	4090.	37826.	38189.	98309.	136497.	60120.
201 MX	43592.	1635420.	129377.	1641109.	43387.	1684495.	-1597722.
202 FZ	3456.	4022.	34678.	35082.	90155.	125237.	55074.
202 MX	23884.	32968.	128539.	134832.	89376.	224208.	-45457.

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY MID, 50T LD UP / SSE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	1391.	453119.	465.	453121.	0.	453121.	-453121.
101 FZ	7768.	59916.	69900.	92392.	102034.	194425.	9642.
101 MX	133853.	30263763.	790092.	30274371.	44267.	30318638.	-30230103.
102 FZ	6135.	59099.	63785.	87172.	93881.	181052.	6709.
102 MX	107256.	1280471.	634667.	1433148.	90260.	1523407.	-1342888.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	61943.	1805.	15239.	63815.	0.	63815.	-63815.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	2003.	414417.	12443.	414608.	-0.	414608.	-414608.
201 FZ	5911.	52090.	67532.	85492.	98309.	183801.	12816.
201 MX	85953.	27976091.	231842.	27977184.	43387.	28020571.	-27933797.
202 FZ	6225.	51890.	61939.	81042.	90155.	171198.	9113.
202 MX	48493.	408205.	229821.	470957.	89376.	560333.	-381582.

(16, 17, 18)

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1. 在 1952 年 10 月 1 日以前

2. 在 1952 年 10 月 1 日以后

3. 在 1952 年 10 月 1 日以前

79604/MJM/AEP, DCC, OLD MAIN TROLLEY MID, 50T LD UP / SSE 'M'

REACTION SUMMARY

LOAD STEP NODE LABEL	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	1391.	33726.	465.	33757.	0.	33757.	-33757.
101 FZ	7768.	6989.	6990.	70676.	102034.	172710.	31358.
101 MX	133853.	2253546.	790092.	2391784.	44267.	2436051.	-2347517.
102 FZ	6135.	6303.	63785.	64388.	93881.	158269.	29492.
102 MX	107256.	112211.	634667.	653374.	90260.	743634.	-563115.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	61943.	1772.	15239.	63814.	0.	63814.	-63814.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	2003.	30856.	12443.	33330.	-0.	33330.	-33330.
201 FZ	5911.	6255.	67532.	68078.	98309.	166387.	30230.
201 MX	85953.	2082353.	231842.	2096982.	43387.	2140368.	-2053595.
202 FZ	6225.	6128.	61939.	62552.	90155.	152708.	27603.
202 MX	48493.	58697.	229821.	242104.	89376.	331480.	-152729.

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY MID, 50T LD DN / OBE \*\*\*

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	651.	241648.	260.	241647.	0.	241647.	-241647.
101 FZ	4477.	31841.	90341.	95892.	102034.	197926.	6142.
101 MX	61144.	16139510.	70410.	16139780.	44267.	16184047.	-16095513.
102 FZ	4271.	31417.	86769.	92380.	93881.	186261.	1500.
102 MX	49754.	682897.	8679.	684762.	90260.	775022.	-594503.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	32357.	857.	9995.	33876.	0.	33876.	-33876.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	915.	221006.	454.	221009.	-0.	221009.	-221009.
201 FZ	3969.	27673.	88798.	93094.	98309.	191403.	5214.
201 MX	37625.	14919501.	54746.	14919649.	43387.	14963036.	-14876262.
202 FZ	3770.	27549.	85432.	89843.	90155.	179998.	313.
202 MX	24275.	217237.	15161.	219114.	89376.	308489.	-129738.

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY MID, 50T LD DN / DRE W

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	651.	34316.	260.	34323.	0.	34323.	-34323
101 FZ	4477.	4548.	90341.	90566.	102034.	192600.	11468
101 MX	61144.	2292296.	70410.	2294192.	44267.	2338459.	-2249924
102 FZ	4271.	4500.	86769.	86991.	93881.	180871.	6890
102 MX	49754.	102114.	8679.	113921.	90260.	204180.	-23661
123 FY	0.	0.	0.	0.	0.	0.	0
123 FZ	0.	0.	0.	0.	0.	0.	0
123 MX	0.	0.	0.	0.	0.	0.	0
123 MY	0.	0.	0.	0.	0.	0.	0
124 FX	32357.	838.	9995.	33876.	0.	33876.	-33876
124 FY	0.	0.	0.	0.	0.	0.	0
124 FZ	0.	0.	0.	0.	0.	0.	0
124 MX	0.	0.	0.	0.	0.	0.	0
124 MY	0.	0.	0.	0.	0.	0.	0
124 MZ	0.	0.	0.	0.	0.	0.	0
201 FY	915.	31390.	454.	31406.	-0.	31406.	-31406
201 FZ	3969.	3956.	88798.	88974.	98309.	187283.	9334
201 MX	37625.	2118750.	54746.	2119791.	43387.	2163178.	-2076404
202 FZ	3770.	3958.	85432.	85606.	90155.	175762.	4549
202 MX	24275.	38263.	15161.	47783.	89376.	137158.	41593

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY MID, 50T LD DN / SSE *W*

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	1294.	453119.	540.	453121.	0.	453121.	-453121.
101 FZ	8573.	59707.	169533.	179944.	102034.	281978.	-77911.
101 MX	114217.	30263766.	136583.	30264290.	44267.	30308557.	-30220023.
102 FZ	8128.	58914.	162765.	173290.	93881.	267171.	-79409.
102 MX	93966.	1280460.	20144.	1284062.	90260.	1374321.	-1193802.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	61613.	1600.	19038.	64507.	0.	64507.	-64507.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	1693.	414417.	881.	414421.	-0.	414421.	-414421.
201 FZ	7603.	51894.	166625.	174684.	98309.	272993.	-76376.
201 MX	76265.	27976094.	107437.	27976404.	43387.	28019791.	-27933017.
202 FZ	7167.	51662.	160244.	168519.	90155.	258674.	-78353.
202 MX	50317.	408174.	32774.	412568.	89376.	501943.	-323192.

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY MID, 50T LD DN / SSE 1X

REACTION SUMMARY

LOAD STEP NODE LABEL	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	1294.	48856.	540.	48876.	0.	48876.	-48876.
101 FZ	8573.	6518.	169533.	169875.	102034.	271909.	-67841.
101 MX	114217.	3263723.	136583.	3268576.	44267.	3312844.	-3224307.
102 FZ	8128.	6480.	162755.	163097.	93881.	256977.	-69216.
102 MX	93966.	150078.	20144.	178210.	90260.	268470.	-87951.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	61613.	1563.	19038.	64506.	0.	64506.	-64506.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	1693.	44690.	881.	44731.	-0.	44731.	-44731.
201 FZ	7603.	5686.	166625.	166895.	98309.	265204.	-68587.
201 MX	76265.	3016442.	107437.	3019318.	43387.	3062705.	-2975931.
202 FZ	7167.	5730.	160244.	160507.	90155.	250662.	-70351.
202 MX	50317.	66482.	32774.	89587.	89376.	178962.	-211.

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY 1/4, 50T LD UP / OBE W

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	7639.	299492.	501.	299590.	0.	299590.	-299590.
101 FZ	7662.	34427.	23834.	42567.	134245.	176912.	91678.
101 MX	511641.	20048028.	63943.	20054657.	43197.	20097854.	-20011461.
102 FZ	7936.	34825.	21176.	41523.	122672.	164196.	81149.
102 MX	120741.	290318.	19220.	315012.	39899.	404911.	-225113.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	21338.	11126.	14092.	27887.	0.	27887.	-27887.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	3995.	140492.	614.	140550.	-0.	140550.	-140550.
201 FZ	3594.	17565.	11086.	21080.	66138.	87218.	45058.
201 MX	243707.	8123637.	30707.	8127350.	39328.	8166678.	-8088022.
202 FZ	3382.	17823.	10321.	20871.	61323.	82195.	40452.
202 MX	66268.	725081.	17377.	728311.	85826.	814137.	-642484.

(lb, in lb)

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY 1/4, 50T LD UP / ONE W

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	7639.	31117.	501.	32045.	0.	32045.	-32045.
101 FZ	7662.	3551.	23834.	25286.	134245.	159531.	108960
101 MX	511641.	2093877.	63943.	2156429.	43197.	2199625.	-2113232
102 FZ	7936.	3572.	21176.	22095.	122672.	145587.	99777
102 MX	120741.	266132.	19220.	292872.	69899.	382771.	-202973
123 FY	0.	0.	0.	0.	0.	0.	0
123 FZ	0.	0.	0.	0.	0.	0.	0
123 MX	0.	0.	0.	0.	0.	0.	0
123 MY	0.	0.	0.	0.	0.	0.	0
124 FX	21338.	2511.	14092.	25695.	0.	25695.	-25695
124 FY	0.	0.	0.	0.	0.	0.	0
124 FZ	0.	0.	0.	0.	0.	0.	0
124 MX	0.	0.	0.	0.	0.	0.	0
124 MY	0.	0.	0.	0.	0.	0.	0
124 MZ	0.	0.	0.	0.	0.	0.	0
201 FY	3995.	14717.	614.	15262.	-0.	15262.	-15262
201 FZ	3594.	1800.	11086.	11793.	66138.	77931.	54345
201 MX	243707.	891330.	30707.	924557.	39328.	963834.	-985229
202 FZ	3382.	1846.	10321.	11017.	61323.	72340.	50307
202 MX	66268.	222331.	17377.	232646.	85826.	318473.	-146820

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY 1/4, 50T LD UP / SSE W

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	14304.	581578.	963.	581761.	0.	581761.	-581761.
101 FZ	18110.	64554.	56785.	87991.	134245.	222237.	46254.
101 MX	959565.	37591906.	146938.	37604438.	43197.	37647635.	-37561241.
102 FZ	17785.	55302.	50510.	84511.	122672.	207183.	38161.
102 MX	226936.	519789.	39625.	568552.	87899.	658451.	-478652.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	39750.	20819.	32036.	55603.	0.	55603.	-55603.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	7502.	263436.	1341.	263546.	-0.	263546.	-263546.
201 FZ	8494.	32936.	26508.	43123.	66138.	109261.	23015.
201 MX	457280.	15231703.	67513.	15238716.	39328.	15278043.	-15199388.
202 FZ	7942.	33420.	24675.	42295.	61323.	103618.	19029.
202 MX	122815.	1354101.	38767.	1360212.	85826.	1446038.	-1274386.

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY 1/4, 50T LD UP / SSE 4"

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	14304.	39642.	963.	42155.	0.	42155.	-42155.
101 FZ	18110.	4494.	56985.	59762.	134245.	174208.	74283.
101 MX	959565.	2680947.	146938.	2351286.	43197.	2894482.	-2908089.
102 FZ	17785.	4508.	50310.	53833.	122672.	176505.	68839.
102 MX	226936.	471786.	39625.	525026.	89899.	614925.	-435127.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	39750.	4236.	32836.	51732.	0.	51732.	-51732.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	7502.	18918.	1341.	20395.	-0.	20395.	-20395.
201 FZ	0494.	2264.	26508.	27928.	66138.	94066.	38210.
201 MX	487280.	1170591.	67513.	1277173.	39328.	1316501.	-1237845.
202 FZ	7942.	2346.	24675.	26027.	61323.	87351.	35296.
202 MX	122815.	386047.	33767.	406963.	85826.	492789.	-321137.

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY 1/4, 50T LD DN / ORE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	7679.	299491.	2515.	299600.	0.	299600.	-299600
101 FZ	12496.	34355.	107729.	113763.	134245.	248008.	20483
101 MX	528054.	20047981.	268216.	20056728.	43197.	20099725.	-20013531
102 FZ	12593.	34895.	104054.	110470.	122672.	233142.	12202
102 MX	79323.	290333.	60598.	307014.	89899.	376913.	-217114
123 FY	0.	0.	0.	0.	0.	0.	0
123 FZ	0.	0.	0.	0.	0.	0.	0
123 MX	0.	0.	0.	0.	0.	0.	0
123 MY	0.	0.	0.	0.	0.	0.	0
124 FX	28272.	11122.	20950.	36903.	0.	36903.	-36903
124 FY	0.	0.	0.	0.	0.	0.	0
124 FZ	0.	0.	0.	0.	0.	0.	0
124 MX	0.	0.	0.	0.	0.	0.	0
124 MY	0.	0.	0.	0.	0.	0.	0
124 MZ	0.	0.	0.	0.	0.	0.	0
201 FY	3761.	140492.	1974.	140556.	-0.	140556.	-140556
201 FZ	4863.	17540.	40712.	44596.	66138.	110734.	21542
201 MX	221374.	8123610.	62897.	8126869.	39328.	8166197.	-8087542
202 FZ	4644.	17848.	39518.	43700.	61323.	105024.	17623
202 MX	75331.	725089.	88543.	734349.	85826.	820175.	-648522

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SRSS-4.3 WHITING CORPORATION ANSYS SRSS PROGRAM  
 TABLE # 032 LS 2 MODE 1 SCALE FACTOR = .1436

79604

87/06/15.

BY MJM PAGE 0-22 OF 58

79604/MJM/AEP, DCC, OLD MAIN TROLLEY 1/4, 50T LD DN / ONE

REACTION SUMMARY

LOAD STEP NODE LABEL	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	7679.	43325.	2516.	44072.	0.	44072.	-44072.
101 FZ	12496.	4949.	107729.	108564.	134245.	242810.	25681.
101 MX	528054.	2907915.	268216.	2967617.	43197.	3010813.	-2924420.
102 FZ	12593.	5025.	104054.	104934.	122672.	227606.	17738.
102 MX	79323.	266407.	60598.	284494.	87899.	374394.	-194595
123 FY	0.	0.	0.	0.	0.	0.	0
123 FZ	0.	0.	0.	0.	0.	0.	0
123 MX	0.	0.	0.	0.	0.	0.	0
123 MY	0.	0.	0.	0.	0.	0.	0
124 FX	28272.	2687.	20950.	35290.	0.	35290.	-35290
124 FY	0.	0.	0.	0.	0.	0.	0
124 FZ	0.	0.	0.	0.	0.	0.	0
124 MX	0.	0.	0.	0.	0.	0.	0
124 MY	0.	0.	0.	0.	0.	0.	0
124 MZ	0.	0.	0.	0.	0.	0.	0
201 FY	3761.	20409.	1974.	20847.	-0.	20847.	-20847
201 FZ	4863.	2526.	40712.	41079.	66138.	107217.	25059
201 MX	221374.	1209219.	62897.	1230923.	39328.	1270251.	-1191596
202 FZ	4644.	2573.	39618.	39973.	61323.	101296.	21351
202 MX	75331.	233062.	88543.	260447.	85826.	346273.	-174620

(lb, in lb)





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79604/MJM/AEP, DCC, OLD MAIN TROLLEY 1/4, 50T LD DN / SSE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	14505.	561577.	5841.	561795.	0.	561795.	-561795.
101 FZ	24181.	64420.	202145.	213535.	134245.	347781.	-79290.
101 MX	1009399.	37591820.	618523.	37610456.	43197.	37653653.	-37567260.
102 FZ	23821.	65435.	195174.	207224.	127672.	329896.	-84552.
102 MX	150887.	519799.	142351.	559662.	89899.	649561.	-469760.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	52505.	20812.	40831.	69693.	0.	69693.	-69693.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	7208.	263436.	4629.	263575.	-0.	263575.	-263575.
201 FZ	9857.	32889.	76509.	83860.	66138.	149978.	-17722.
201 MX	413484.	15231654.	131519.	15237833.	39328.	15277160.	-15198505.
202 FZ	-9308.	33466.	74420.	82128.	61323.	143451.	-20807.
202 MX	154894.	1354108.	207965.	1378713.	85826.	1464539.	-1292886.

(lb, in lb)

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY 1/4, 50T LD DN / SSE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	14505.	61028.	5841.	63000.	0.	63000.	-63000.
101 FZ	24181.	6962.	202145.	203705.	134245.	337951.	-69460
101 MX	1009399.	4102855.	618523.	4270231.	43197.	4313428	-4227035
102 FZ	23821.	7063.	195174.	196749.	122672.	319421.	-74077
102 MX	150887.	472144.	142351.	515704.	89899.	605604.	-425805
123 FY	0.	0.	0.	0.	0.	0.	0
123 FZ	0.	0.	0.	0.	0.	0.	0
123 MX	0.	0.	0.	0.	0.	0.	0
123 MY	0.	0.	0.	0.	0.	0.	0
124 FX	52505.	4450.	40831.	66662.	0.	66662.	-66662
124 FY	0.	0.	0.	0.	0.	0.	0
124 FZ	0.	0.	0.	0.	0.	0.	0
124 MX	0.	0.	0.	0.	0.	0.	C
124 MY	0.	0.	0.	0.	0.	0.	C
124 MZ	0.	0.	0.	0.	0.	0.	C
201 FY	7208.	28838.	4629.	30083.	-0.	30083.	-30083
201 FZ	9857.	3552.	76509.	77223.	66138.	143361.	-11085
201 MX	413484.	1731876.	131519.	1785402.	39328.	1824730.	-1746075
202 FZ	9308.	3617.	74420.	75087.	61323.	136410.	-13760
202 MX	154894.	400749.	207965.	477328.	85826.	563154.	-391501

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY END, 50T LD UP / ORE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	9458.	124413.	1517.	124781.	0.	124781.	-124781
101 FZ	8958.	10367.	14579.	20008	148423.	168430.	128415
101 MX	566820.	7521156.	114640.	7543356.	41630.	7584986.	-7501726
102 FZ	8617.	12098.	12447.	19379.	137227.	156608.	117850
102 MX	92425.	1297616.	17988.	1303023.	70937.	1373762.	-1212084
123 FY	0.	0.	0.	0.	0.	0.	0
123 FZ	0.	0.	0.	0.	0.	0.	0
123 MX	0.	0.	0.	0.	0.	0.	0
123 MY	0.	0.	0.	0.	0.	0.	0
124 FX	22763.	13537.	15102.	30488.	0.	30488.	-30488
124 FY	0.	0.	0.	0.	0.	0.	0
124 FZ	0.	0.	0.	0.	0.	0.	0
124 MX	0.	0.	0.	0.	0.	0.	0
124 MY	0.	0.	0.	0.	0.	0.	0
124 MZ	0.	0.	0.	0.	0.	0.	0
201 FY	4321.	55248.	619.	55421.	-0.	55421.	-55421
201 FZ	3946.	5791.	5903.	9163.	51973.	61136.	42810
201 MX	171249.	2166056.	40011.	2173184.	37940.	2211123.	-2135244
202 FZ	3800.	6597.	5193.	9216.	46754.	55970.	37538
202 MX	32874.	706258.	12112.	707127.	85033.	792160.	-622090

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY END, 50T LD UP, / ONE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
NODE LABEL							
101 FY	9458.	38364.	1517.	39542.	0.	39542.	-39542.
101 FZ	8958.	3118.	14579.	17393.	148423.	165815.	131030.
101 MX	566820.	2611714.	114640.	2674972.	41630.	2716603.	-2633342.
102 FZ	8617.	3757.	12447.	15598.	137229.	152827.	121631.
102 MX	92425.	795582.	17988.	801135.	90939.	892074.	-710196.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	22763.	3330.	15102.	27519.	0.	27519.	-27519.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	4321.	19067.	619.	19560.	-0.	19560.	-19560.
201 FZ	3946.	1695.	5903.	7300.	51973.	59273.	44673.
201 MX	171249.	1305205.	40011.	1317000.	37940.	1354939.	-1279060.
202 FZ	3800.	1758.	5193.	6671.	46754.	53425.	40083.
202 MX	32874.	704022.	12112.	704893.	85033.	789926.	-619859.

(lb, in lb)





79604/MJM/AEP, DCC, OLD MAIN TROLLEY END, 50T LD UP / SSE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	18148.	238174.	3154.	238085.	0.	238805.	-238805.
101 FZ	20903.	19969.	34857.	45285.	148423.	193707.	103138.
101 MX	1090343.	14392910.	251470.	14436342.	41630.	14477972.	-14394712.
102 FY	19214.	23349.	29689.	42377.	137229.	179606.	94052.
102 MX	177749.	2453840.	36569.	2460542.	90939.	2551481.	-2369602.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	43037.	25955.	35050.	61273.	0.	61273.	-61273.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	8308.	105576.	1184.	105909.	-0.	105909.	-105909.
201 FZ	8926.	11079.	14115.	20041.	51973.	72014.	31932.
201 MX	330448.	4089399.	82349.	4103555.	37940.	4141475.	-4065615.
202 FY	8294.	12629.	12396.	19543.	46754.	66297.	27211.
202 MX	63933.	1295153.	26002.	1296991.	85033.	1382024.	-1211957.

(lb, in lb)



79604/MJM/AEP, DCC, OLD MAIN TROLLEY END, 50T LD UP / SSE W

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	18148.	59451.	3154.	62240.	0.	62240.	-62240.
101 FZ	20903.	5228.	34857.	40779.	148423.	189402.	107444.
101 MX	1090343.	4270530.	251490.	4414694.	41630.	4456324.	-4373064.
102 FZ	19214.	6551.	29689.	35966.	137229.	173195.	101263.
102 MX	177749.	1418050.	36569.	1429615.	90939.	1520554.	-1338675.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	43037.	4454.	35050.	55682.	0.	55682.	-55682.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	8308.	30580.	1184.	31711.	-0.	31711.	-31711.
201 FZ	8926.	2509.	14115.	16888.	51973.	68861.	35035.
201 MX	330448.	2318412.	82349.	2343291.	37940.	2381231.	-2305351.
202 FZ	8294.	2438.	12396.	15112.	46754.	61866.	31642.
202 MX	63933.	1290520.	26002.	1292365.	85033.	1377398.	-1207331.

(lb, in lb)



79604/MJM/AEP, DCC, OLD MAIN TROLLEY END, 50T LD DN / OKE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	9485.	124404.	1554.	124775.	0.	124775.	-124775.
101 FZ	11859.	12266.	100964.	102395.	140423.	250818.	46027.
101 MX	570619.	7520127.	115524.	7542630.	41630.	7584260.	-7501000.
102 FZ	11725.	12047.	99231.	100645.	137229.	237874.	36584.
102 MX	86981.	1299553.	12447.	1302520.	90939.	1393459.	-1211581.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	24785.	13494.	17575.	33245.	0.	33245.	-33245.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	4222.	55248.	568.	55412.	-0.	55412.	-55412.
201 FZ	4730.	6172.	19398.	20899.	51973.	72872.	31074.
201 MX	146836.	2166144.	14328.	2171163.	37940.	2209102.	-2133223.
202 FZ	4431.	6589.	18786.	20395.	46754.	67149.	26359.
202 MX	24446.	706144.	10741.	706649.	65033.	791682.	-621616.

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY END, 50T LD DN / UBE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	9485.	48861.	1554.	49797.	0.	49797.	-49797.
101 FZ	11859.	5083.	100964.	101785.	148423.	250208.	46637.
101 MX	570619.	3172331.	115524.	3225311.	41630.	3266941.	-3183681.
102 FZ	11725.	5294.	99231.	100062.	137229.	237291.	37168.
102 MX	86981.	837853.	12447.	842448.	90939.	933337.	-751509.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	24785.	4765.	17575.	30755.	0.	30755.	-30755.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	4222.	23233.	568.	23620.	-0.	23620.	-23620.
201 FZ	4730.	2451.	19398.	20117.	51973.	72090.	31856.
201 MX	146836.	1378161.	14328.	1386036.	37940.	1423975.	-1348096.
202 FZ	4431.	2517.	18786.	19465.	46754.	66219.	27289.
202 MX	24446.	704054.	10741.	704560.	85033.	789594.	-619527.

(lb) in lb)



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79604/MJM/AEP, DCC, OLD MAIN TROLLEY END, 50T LD DN / SSE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	18254.	238150.	3357.	238872.	0.	238872.	-238872.
101 FZ	23256.	23568.	189387.	192259.	148423.	340682.	-43837.
101 MX	1105214.	14390159.	263896.	14434951.	41630.	14476581.	-14393321.
102 FZ	22179.	23226.	186073.	188824.	137229.	326053.	-51595.
102 MX	167216.	2453678.	27237.	2459520.	90939.	2550459.	-2368581.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	47263.	26010.	35459.	64557.	0.	64557.	-64557.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	8108.	105575.	1033.	105892.	-0.	105892.	-105892.
201 FZ	10032.	11847.	36619.	39774.	51973.	91747.	12199.
201 MX	281493.	4089592.	32768.	4099399.	37940.	4137339.	-4061459.
202 FZ	9131.	12633.	35401.	38681.	46754.	85435.	8074.
202 MX	48478.	1294846.	24556.	1295986.	85033.	1381019.	-1210953.

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CONFIDENTIAL



SRSS-4.3 WHITING CORPORATION ANSYS SRSS PROGRAM  
 TABLE # 042 LS 2 NODE 2 SCALE FACTOR = .2385

79604 87/06/16.  
 BY MJM PAGE 0-32 OF 58  
 001127687

79604/MJM/AEP, DCC, OLD MAIN TROLLEY END, 50T LD DN / GSE W

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
NODE LABEL							
101 FY	18254.	74899.	3357.	77164.	0.	77164.	-77164
101 FZ	23256.	8111.	189387.	190782.	148423.	339404.	-42559
101 MX	1105214.	5056053.	263876.	5182163.	41630.	5223793.	-5140533
102 FZ	22179.	8908.	186073.	187602.	137229.	324831.	-50373
102 MX	167216.	1472227.	27237.	1481943.	90939.	1572882.	-1391004
123 FY	0.	0.	0.	0.	0.	0.	0
123 FZ	0.	0.	0.	0.	0.	0.	0
123 MX	0.	0.	0.	0.	0.	0.	0
123 MY	0.	0.	0.	0.	0.	0.	0
124 FX	47263.	7327.	35459.	59538.	0.	59538.	-59538
124 FY	0.	0.	0.	0.	0.	0.	0
124 FZ	0.	0.	0.	0.	0.	0.	0
124 MX	0.	0.	0.	0.	0.	0.	0
124 MY	0.	0.	0.	0.	0.	0.	0
124 MZ	0.	0.	0.	0.	0.	0.	0
201 FY	8108.	36548.	1083.	37452.	-0.	37452.	-37452
201 FZ	10032.	3841.	36619.	38162.	51973.	90135.	13811
201 MX	281493.	2412508.	32768.	2429096.	37940.	2467036.	-2391157
202 FZ	9131.	3807.	35401.	36757.	46754.	83511.	9997
202 MX	48478.	1290395.	24556.	1291539.	85033.	1376572.	-1206505

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY\_MID, NO LD / OBE *111*

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
NODE LABEL							
101 FY	651.	241646.	174.	241647.	0.	241647.	-241647.
101 FZ	2818.	31807.	13366.	34616.	77038.	111654.	42421.
101 MX	61465.	16139502.	32461.	16139652.	43418.	16183070.	-16096234.
102 FZ	2597.	31449.	11463.	33574.	68869.	102443.	35295.
102 MX	50387.	682901.	12219.	684867.	89442.	774308.	-595425.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	32423.	854.	4497.	32744.	0.	32744.	-32744.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	924.	221007.	229.	221009.	-0.	221009.	-221009.
201 FZ	2006.	27639.	12811.	30529.	73312.	103842.	42783.
201 MX	37572.	14919495.	28612.	14919569.	42515.	14962085.	-14877054.
202 FZ	1778.	27582.	10829.	29684.	65144.	94828.	35459.
202 MX	23751.	217251.	11012.	218822.	88536.	307358.	-130287.

(16, in 16)

THE UNIVERSITY OF CHICAGO

1964

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SRSS-4.3 WHITING CORPORATION ANSYS SRSS PROGRAM  
 TABLE # 844 LS 2 MODE 1 SCALE FACTOR = .0801

79604

87/06/17.

BY MJM

PAGE 8-34 OF 58

79604/MJM/AEP, DCC, OLD MAIN TROLLEY MID, NO LD / OBE 'X'

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	651.	19360.	174.	19372.	0.	19372.	-19372.
101 FZ	2818.	2603.	13366.	13906.	77038.	90943.	63132.
101 MX	61465.	1293655.	32461.	1295521.	43418.	1338939.	-1252104.
102 FZ	2597.	2575.	11463.	12032.	68869.	80902.	56837.
102 MX	50387.	63514.	12219.	81989.	89442.	171431.	7453.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	32423.	834.	4497.	32744.	0.	32744.	-32744.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	924.	17715.	229.	17740.	-0.	17740.	-17740.
201 FZ	2006.	2262.	12811.	13163.	73312.	86475.	60150.
201 MX	37572.	1195383.	28612.	1196315.	42515.	1238830.	-1153800.
202 FZ	1778.	2283.	10829.	11209.	65144.	76352.	53935.
202 MX	23751.	28769.	11012.	38898.	88536.	127433.	49638.

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY MID, NO LD / SSE *W*

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
NODE LABEL							
101 FY	1287.	453119.	352.	453121.	0.	453121.	-453121.
101 FZ	5987.	59644.	31962.	67933.	77038.	144970.	9105.
101 MX	114435.	30263751.	74935.	30264060.	43418.	30307478.	-30220643.
102 FZ	5451.	58975.	27412.	65262.	68869.	134132.	3607.
102 MX	95150.	1280471.	27069.	1284287.	89442.	1373728.	-1194845.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	61748.	1580.	10525.	62658.	0.	62658.	-62658.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	1709.	414417.	462.	414421.	-0.	414421.	-414421.
201 FZ	4598.	51830.	30635.	60382.	73312.	133695.	12930.
201 MX	75797.	27976082.	66882.	27976265.	42515.	28018780.	-27933750.
202 FZ	4024.	51724.	25894.	57984.	65144.	123127.	7160.
202 MX	48954.	408215.	25947.	411958.	88536.	500494.	-323423.

(lb, in lb)

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SRSS-4.3 WHITING CORPORATION ANSYS SRSS PROGRAM  
 TABLE # 846 LS 2 MODE-1 SCALE FACTOR = .0554

79604 87/06/17.  
 BY M.J.M. PAGE 8-36 OF 58  
*R. H. & 2687*

79604/MJM/AEP, DCC, OLD MAIN TROLLEY MID, NO LD / SSE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	1287.	25121.	352.	25157.	0.	25157.	-25157.
101 FZ	5987.	3491.	31962.	32705.	77038.	109742.	44333.
101 MX	114435.	1679176.	74935.	1684738.	43418.	1728156.	-1641320.
102 FZ	5451.	3472.	27412.	28164.	68869.	97033.	40705.
102 MX	95150.	92480.	27069.	135421.	89442.	224862.	-45979.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	61748.	1542.	10525.	62657.	0.	62657.	-62657.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	1709.	22990.	462.	23058.	-0.	23058.	-23058.
201 FZ	4598.	3061.	30635.	31129.	73312.	104441.	42184.
201 MX	75797.	1551117.	66882.	1554407.	42515.	1596923.	-1511892.
202 FZ	4024.	3143.	25894.	26393.	65144.	91537.	38751.
202 MX	48954.	55198.	25947.	78208.	88536.	166744.	10327.

(16, in 16)

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY 1/4, NO LD / OBE *W*

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	7570.	299492.	700.	299588.	0.	299588.	-299588.
101 FZ	7539.	34459.	11498.	37101.	96753.	133854.	59653.
101 MX	509939.	20048044.	78267.	20054681.	41923.	20096603.	-20012758.
102 FZ	7942.	34792.	9142.	36840.	85152.	121992.	48312.
102 MX	74318.	290332.	21240.	300445.	88685.	389130.	-211760.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	29151.	11113.	12583.	33639.	0.	33639.	-33639.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	3624.	140492.	526.	140540.	-0.	140540.	-140540.
201 FZ	3225.	17577.	6260.	18935.	53637.	72573.	34702.
201 MX	220891.	8123646.	15609.	8126663.	38847.	8165511.	-8087816.
202 FZ	2894.	17811.	5653.	18909.	48820.	67729.	29911.
202 MX	59766.	725083.	28522.	728101.	85403.	813504.	-642699.

(lb, in lb)

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY 1/4, NO LD / OBE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	7570.	23803.	700.	24988.	0.	24988.	-24988.
101 FZ	7539.	2698.	11498.	14011.	96753.	110765.	82742.
101 MX	509939.	1607676.	78267.	1688427.	41923.	1730349.	-1646504.
102 FZ	7942.	2703.	9142.	12408.	85152.	97560.	72744.
102 MX	74318.	266032.	21240.	277033.	88685.	365718.	-188348.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	29151.	2364.	12583.	31838.	0.	31838.	-31838.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	3624.	11323.	526.	11901.	-0.	11901.	-11901.
201 FZ	3225.	1371.	6260.	7174.	53637.	60811.	46464.
201 MX	220891.	706609.	15609.	740495.	38847.	779342.	-701647.
202 FZ	2894.	1400.	5653.	6503.	48820.	55323.	42317.
202 MX	59766.	217437.	28522.	227297.	85403.	312700.	-141895.

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY 1/4, NO LD / SSE

REACTION SUMMARY

LOAD STEP NODE LABEL	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	14177.	561578.	1597.	561759.	0.	561759.	-561759.
101 FZ	16636.	64616.	27501.	72169.	96753.	168922.	24585.
101 MX	955632.	37591937.	183870.	37604531.	41923.	37646454.	-37562609.
102 FZ	16309.	65241.	21783.	70688.	85152.	155840.	14464.
102 MX	133585.	519798.	49478.	538965.	88685.	627650.	-450280.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	54272.	20795.	28306.	64646.	0.	64646.	-64646.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	6788.	263436.	1214.	263526.	-0.	263526.	-263526.
201 FZ	7768.	32960.	15001.	37037.	53637.	90674.	16600
201 MX	411454.	15231720.	33352.	15237313.	38847.	15276160.	-15198466
202 FZ	6990.	33397.	13547.	36711.	48820.	85531.	12109
202 MX	108988.	1354097.	67500.	1360152.	85403.	1445554.	-1274749

(lb) in lb)

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SRSS-4.3 WHITING CORPORATION ANSYS SRSS PROGRAM  
 TABLE # 850 LS 2 MODE 1 SCALE FACTOR = .0521

79604 87/06/17.  
 BY MJM PAGE 8-40 OF 58  
 79604

79604/MJM/AEP, DCC, OLD MAIN TROLLEY 1/4, NO LD / SSE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	14177.	30718.	1597.	33869.	0.	33869.	-33869.
101 FZ	16636.	3451.	27501.	32326.	96753.	129080.	64427.
101 MX	955632.	2091436.	183870.	2306760.	41923.	2348683.	-2264838.
102 FZ	16309.	3418.	21783.	27426.	85152.	112578.	57726.
102 MX	133585.	471695.	49478.	492736.	88685.	581421.	-404051.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	54272.	4053.	28306.	61344.	0.	61344.	-61344.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	6788.	14825.	1214.	16350.	-0.	16350.	-16350.
201 FZ	7768.	1746.	15001.	16983.	53637.	70621.	36654.
201 MX	411454.	977787.	33352.	1061355.	38847.	1100202.	-1022508.
202 FZ	6990.	1788.	13547.	15348.	48820.	64168.	33472.
202 MX	108988.	381647.	67500.	402602.	85403.	488003.	-317200.

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY RH END, NO LD / OBE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	4723.	52689.	918.	52908.	0.	52908.	-52908.
101 FZ	6374.	3301.	5745.	9194.	107363.	116557.	98169.
101 MX	242849.	3188424.	69730.	3198419.	39266.	3237685.	-3159153.
102 FZ	6698.	4182.	4369.	9025.	96327.	105352.	87302.
102 MX	127156.	1210645.	19009.	1217453.	91092.	1308545.	-1126361.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	24162.	6711.	12596.	28063.	0.	28063.	-28063.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	2386.	29456.	330.	29554.	-0.	29554.	-29554.
201 FZ	4180.	2463.	3629.	6059.	43041.	49100.	36982.
201 MX	75526.	1616605.	20677.	1618500.	37117.	1655617.	-1581383.
202 FZ	4195.	2848.	3067.	5926.	37631.	43557.	31706.
202 MX	50351.	901107.	7424.	902543.	84353.	986896.	-818190.

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SRSS-4.3 WHITING CORPORATION ANSYS SRSS PROGRAM  
 TABLE # 852 LS 2 MODE 1 SCALE FACTOR = 3696

79604 87/06/17.  
 BY MJM PAGE 42 OF 58  
*REV 8/88*

79604/MJM/AEP, DCC, OLD MAIN TROLLEY RH END, NO LD / OBE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
NODE LABEL							
101 FY	4723.	43606.	918.	43871.	0.	43871.	-43871.
101 FZ	6374.	2755.	5745.	9013.	107363.	116376.	98351.
101 MX	242849.	2919399.	69730.	2930312.	39266.	2969578.	-289104.
102 FZ	6698.	3285.	4369.	8646.	96327.	104973.	87687.
102 MX	127156.	822455.	19009.	832444.	91092.	923536.	-741352.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	24162.	3637.	12596.	27490.	0.	27490.	-27490.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	2386.	25550.	330.	25664.	-0.	25664.	-25664.
201 FZ	4180.	2078.	3629.	5913.	43041.	48954.	37121.
201 MX	75526.	1615928.	20677.	1617824.	37117.	1654941.	-1580701.
202 FZ	4195.	2214.	3067.	5649.	37631.	43280.	31981.
202 MX	50351.	855039.	7424.	856552.	84353.	940905.	-772191.

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY RH END, NO LD / SSE \*\*\*

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	9158.	98535.	1900.	98978.	0.	98978.	-98978.
101 FZ	14291.	6626.	12952.	20393.	107363.	127756.	86970.
101 MX	480112.	5921221.	154045.	5942651.	39266.	5981917.	-5903384.
102 FZ	13393.	8315.	9500.	18405.	96327.	114733.	77922.
102 MX	245043.	2283548.	36669.	2296950.	91092.	2388042.	-2205858.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	47007.	12988.	27806.	56139.	0.	56139.	-56139.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	4626.	54352.	622.	54552.	-0.	54552.	-54552.
201 FZ	9479.	4573.	8386.	13457.	43041.	56499.	29584.
201 MX	143124.	2922623.	45843.	2926485.	37117.	2963601.	-2889368.
202 FZ	8909.	5322.	7004.	12520.	37631.	50151.	25111.
202 MX	95390.	1642764.	14628.	1645597.	84353.	1729950.	-1561244.

(lb, in lb)

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SRSS-4.3 WHITING CORPORATION ANSYS SRSS PROGRAM

79604 87/06/17.

TABLE # 057LS 2 MODE 1 SCALE FACTOR = .2156

BY MJM PAGE 0=44 OF 58

79604/MJM/AEP, DCC, OLD MAIN TROLLEY RH END, NO LD / SSE

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	9158.	78204.	1900.	78761.	0.	78761.	-78761
101 FZ	14291.	5508.	12952.	20058.	107363.	127421.	87305
101 MX	480112.	5320727.	154045.	5344565.	39266.	5383831.	-5305299
102 FZ	13393.	6450.	9500.	17641.	96327.	113969.	78686
102 MX	245043.	1404326.	36669.	1426017.	91092.	1517109.	-1334925
123 FY	0.	0.	0.	0.	0.	0.	0
123 FZ	0.	0.	0.	0.	0.	0.	0
123 MX	0.	0.	0.	0.	0.	0.	0
123 MY	0.	0.	0.	0.	0.	0.	0
124 FX	47007.	6167.	27806.	54962.	0.	54962.	-54962
124 FY	0.	0.	0.	0.	0.	0.	0
124 FZ	0.	0.	0.	0.	0.	0.	0
124 MX	0.	0.	0.	0.	0.	0.	0
124 MY	0.	0.	0.	0.	0.	0.	0
124 MZ	0.	0.	0.	0.	0.	0.	0
201 FY	4626.	45514.	622.	45753.	-0.	45753.	-45753
201 FZ	9479.	3705.	8386.	13188.	43041.	56229.	29854
201 MX	143124.	2921086.	43843.	2924949.	37117.	2962066.	-2887832
202 FZ	8909.	3892.	7004.	11982.	37631.	49614.	25649
202 MX	95390.	1538277.	14628.	1541302.	84353.	1625655.	-1456949

(lb, in lb)



TABLE 8-55

## ORE GIRDER TO END TIE CONNECTION

SUM OF ELEMENT FORCES IN ELEMENT CO-ORDINATE SYSTEM (lb, in-lb)

	ELEM	NODE			FX	FY	FZ	MX	MY	MZ	
MID	50	U	14	I	151	16043.	31397.	1883.	7186.	75429.	2660800.
MID	50	U	15	I	155	16043.	30706.	2006.	7186.	81974.	3678000.
MID	50	U	72	I	251	10918.	31453.	1869.	7343.	73412.	3730600.*
MID	50	U	73	I	255	10918.	31489.	1974.	7343.	81149.	2632800.
MID	50	D	14	I	151	19532.	33957.	1919.	7962.*	79934.	3225800.
MID	50	D	15	I	155	19532.	34000.*	2052.	7962.*	85839.	3607200.
MID	50	D	72	I	251	14076.	28447.	1907.	7567.	78280.	3040000.
MID	50	D	73	I	255	14076.	28622.	2038.	7567.	84808.	2693500.
1/4	50	U	14	I	151	20947.	17914.	1894.	3909.	77327.	1598200.
1/4	50	U	15	I	155	20947.	17933.	2013.	3909.	81926.	2035300.
1/4	50	U	72	I	251	7833.	19224.	1843.	3928.	69930.	2121200.
1/4	50	U	73	I	255	7833.	19288.	1901.	3928.	72781.	1809200.
1/4	50	D	14	I	151	28780.	23114.	1952.	4199.	84590.	2101100.
1/4	50	D	15	I	155	28780.	23250.	2078.*	4199.	87149.*	2587800.
1/4	50	D	72	I	251	10000.	26584.	1872.	4803.	73360.	2901000.
1/4	50	D	73	I	255	10000.	26659.	1929.	4803.	75181.	2492100.
END	50	U	14	I	151	29151.	21424.	1920.	2528.	81186.	1537100.
END	50	U	15	I	155	29151.	21182.	2030.	2528.	82415.	2778000.
END	50	U	72	I	251	15569.	22476.	1853.	4828.	72249.	2459100.
END	50	U	73	I	255	15569.	22368.	1914.	4828.	72715.	2631300.
END	50	D	14	I	151	38544.*	21467.	1964.	2528.	86420.	1576900.
END	50	D	15	I	155	38544.*	21272.	2078.*	2528.	86446.	2777500.
END	50	D	72	I	251	15788.	29950.	1870.	5747.	73735.	3545000.
END	50	D	73	I	255	15788.	29990.	1922.	5747.	73771.	2999700.
MID	NO		14	I	151	11066.	19294.	1844.	6215.	70725.	1825300.
MID	NO		15	I	155	11066.	19300.	1956.	6215.	77762.	2061100.
MID	NO		72	I	251	7974.	16318.	1837.	5996.	69804.	1748800.
MID	NO		73	I	255	7974.	16409.	1947.	5996.	77115.	1540900.
1/4	NO		14	I	151	16398.	15056.	1872.	3635.	74327.	1315300.
1/4	NO		15	I	155	16398.	14997.	1965.	3635.	77853.	1737100.
1/4	NO		72	I	251	6594.	14876.	1829.	3401.	68306.	1664800.
1/4	NO		73	I	255	6594.	14925.	1881.	3401.	70985.	1407900.
RHE	NO		14	I	151	32938.	21203.	1940.	1919.	83936.	1443000.
RHE	NO		15	I	155	32938.	21125.	2026.	1919.	81224.	2826200.
RHE	NO		72	I	251	19222.	28387.	1875.	5672.	75245.	2472600.
RHE	NO		73	I	255	19222.	28053.	1930.	5672.	73722.	3531400.

\* DENOTES A MAXIMUM



TABLE B56

## SSE GIRDER TO END TIE CONNECTION

SUM OF ELEMENT FORCES IN ELEMENT CO-ORDINATE SYSTEM (lb, in-lb)

		ELEM NODE			FX	FY	FZ	MX	MY	MZ
MID	50 U	14 I	151	21542.	45516.	1935.	8881.	81438.	3588800.	
MID	50 U	15 I	155	21542.	43946.	2051.	8881.	85805.	5572400.	
MID	50 U	72 I	251	13950.	49060.	1912.	9520.	77996.	5996700.	
MID	50 U	73 I	255	13950.	49034.	2037.	9520.	84829.	3913200.	
MID	50 D	14 I	151	27797.	48384.	2002.	10480.*	89851.	4594700.	
MID	50 D	15 I	155	27797.	48437.	2143.	10480.*	93466.	5140600.	
MID	50 D	72 I	251	20028.	40584.	1985.	10025.	87497.	4347600.	
MID	50 D	73 I	255	20028.	40827.	2123.	10025.	92097.	3832800.	
1/4	50 U	14 I	151	27566.	25853.	1944.	5358.	83623.	2236000.	
1/4	50 U	15 I	155	27566.	25721.	2068.	5358.	86226.	2984700.	
1/4	50 U	72 I	251	11503.	25017.	1867.	4904.	72983.	2818800.	
1/4	50 U	73 I	255	11503.	25114.	1932.	4904.	75105.	2369200.	
1/4	50 D	14 I	151	41197.	34329.	2048.	5769.	96331.	3069900.	
1/4	50 D	15 I	155	41197.	34435.	2182.	5769.	95418.	3877400.	
1/4	50 D	72 I	251	15089.	37878.	1918.	6421.	78868.	4160300.	
1/4	50 D	73 I	255	15089.	37989.	1978.	6421.	79191.	3562500.	
END	50 U	14 I	151	43470.	39246.	2027.	4068.	95340.	2767800.	
END	50 U	15 I	155	43470.	38748.	2150.	4068.	90998.	5099100.	
END	50 U	72 I	251	28267.	32414.	1903.	6617.	80005.	3109400.	
END	50 U	73 I	255	28267.	31987.	1987.	6617.	77414.	4441400.	
END	50 D	14 I	151	57701.*	39292.	2091.	4066.	102910.	2817000.	
END	50 D	15 I	155	57701.*	38856.	2220.*	4066.	97105.	5098100.	
END	50 D	72 I	251	28531.	44051.	1929.	8000.	82034.	4948800.	
END	50 D	73 I	255	28531.	43930.	1998.	8000.	78896.	4942600.	
MID	NO	14 I	151	14376.	25195.	1877.	7889.	74649.	2375300.	
MID	NO	15 I	155	14376.	25173.	1992.	7889.	80787.	2700000.	
MID	NO	72 I	251	10353.	21482.	1867.	7677.	73408.	2320700.	
MID	NO	73 I	255	10353.	21585.	1982.	7677.	80014.	2011400.	
1/4	NO	14 I	151	22264.	22804.	1917.	4909.	79905.	1925400.	
1/4	NO	15 I	155	22264.	22579.	2011.	4909.	81470.	2673300.	
1/4	NO	72 I	251	10117.	19833.	1850.	4244.	70999.	2286800.	
1/4	NO	73 I	255	10117.	19910.	1908.	4244.	72975.	1894500.	
RHE	NO	14 I	151	57245.	37612.	2086.	3089.	103250.*	2578400.	
RHE	NO	15 I	155	57245.	37546.	2192.	3089.	93236.	4987600.	
RHE	NO	72 I	251	34657.	49500.*	1957.	8738.	86433.	3848500.	
RHE	NO	73 I	255	34657.	48769.	2023.	8738.	80161.	6358900.*	

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TABLE 057

## OBE TROLLEY REACTIONS

SUM OF ELEMENT FORCES IN ELEMENT CO-ORDINATE SYSTEM (lb, in-lb)

		ELEM	NODE		FX	FY	FZ	MX	MY	MZ
MID	50	U	60	J 391	111710.	0.	0.	0.	0.	0.
MID	50	U	61	J 393	70629.	18774.	11484.	0.	0.	0.
MID	50	U	62	J 392	111120.	20165.	0.	0.	0.	0.
MID	50	U	63	J 394	71057.	0.	11946.	0.	0.	0.
MID	50	D	60	J 391	161740.	0.	0.	0.	0.	0.
MID	50	D	61	J 393	118900.	21813.	11721.	0.	0.	0.
MID	50	D	62	J 392	161490.	25653.	0.	0.	0.	0.
MID	50	D	63	J 394	119130.	0.	11757.	0.	0.	0.
1/4	50	U	60	J 391	107120.	-0.	0.	0.	0.	0.
1/4	50	U	61	J 393	45590.	10718.	8623.	0.	0.	0.
1/4	50	U	62	J 392	75758.	21635.	0.	0.	0.	0.
1/4	50	U	63	J 394	78111.	0.	8479.	0.	0.	0.
1/4	50	D	60	J 391	161840.*	0.	0.	0.	0.	0.
1/4	50	D	61	J 393	95357.	14473.	11872.	0.	0.	0.
1/4	50	D	62	J 392	126770.	29841.	0.	0.	0.	0.
1/4	50	D	63	J 394	130980.	0.	11815.	0.	0.	0.
END	50	U	60	J 391	102230.	0.	0.	0.	0.	0.
END	50	U	61	J 393	50635.	15205.	11430.	0.	0.	0.
END	50	U	62	J 392	77394.	28127.	0.	0.	0.	0.
END	50	U	63	J 394	76243.	0.	12029.	0.	0.	0.
END	50	D	60	J 391	145890.	0.	0.	0.	0.	0.
END	50	D	61	J 393	92174.	15370.	13332.	0.	0.	0.
END	50	D	62	J 392	118460.	39380.*	0.	0.	0.	0.
END	50	D	63	J 394	120080.	0.	14116.*	0.	0.	0.
MID	NO		60	J 391	63063.	0.	0.	0.	0.	0.
MID	NO		61	J 393	28189.	12423.	10811.	0.	0.	0.
MID	NO		62	J 392	63068.	14541.	0.	0.	0.	0.
MID	NO		63	J 394	28015.	0.	10835.	0.	0.	0.
1/4	NO		60	J 391	74837.	0.	0.	0.	0.	0.
1/4	NO		61	J 393	14259.	8622.	9968.	0.	0.	0.
1/4	NO		62	J 392	47858.	16797.	0.	0.	0.	0.
1/4	NO		63	J 394	46043.	0.	10021.	0.	0.	0.
RHE	NO		60	J 391	74139.	0.	0.	0.	0.	0.
RHE	NO		61	J 393	26048.	15405.	11571.	0.	0.	0.
RHE	NO		62	J 392	53301.	25359.	0.	0.	0.	0.
RHE	NO		63	J 394	49360.	0.	12015.	0.	0.	0.

\* DENOTES A MAXIMUM

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TABLE B58

OBE TROLLEY REACTIONS

DIFFERENCE OF ELEMENT FORCES IN ELEMENT CO-ORDINATE SYSTEM (lb. in-lb)

		ELEM	NODE		FX	FY	FZ	MX	MY	MZ
MID	50	U	60	J 391	38715.	0.	0.	0.	0.	0.
MID	50	U	61	J 393	22955.	-18555.	-11366.	0.	0.	0.
MID	50	U	62	J 392	39311.	-19946.	0.	0.	0.	0.
MID	50	U	63	J 394	22523.	0.	-11828.	0.	0.	0.
MID	50	D	60	J 391	-11312.	0.	0.	0.	0.	0.
MID	50	D	61	J 393	-25317.	-21594.	-11603.	0.	0.	0.
MID	50	D	62	J 392	-11054.	-25434.	0.	0.	0.	0.
MID	50	D	63	J 394	-25553.	0.	-11639.	0.	0.	0.
1/4	50	U	60	J 391	69725.	0.	0.	0.	0.	0.
1/4	50	U	61	J 393	21572.	-10379.	-8440.	0.	0.	0.
1/4	50	U	62	J 392	48253.	-21296.	0.	0.	0.	0.
1/4	50	U	63	J 394	41890.	0.	-8296.	0.	0.	0.
1/4	50	D	60	J 391	15005.	0.	0.	0.	0.	0.
1/4	50	D	61	J 393	-28194.*	-14134.	-11689.	0.	0.	0.
1/4	50	D	62	J 392	-2755.	-29502.	0.	0.	0.	0.
1/4	50	D	63	J 394	-10983.	0.	-11632.	0.	0.	0.
END	50	U	60	J 391	71553.	0.	0.	0.	0.	0.
END	50	U	61	J 393	19589.	-14554.	-11078.	0.	0.	0.
END	50	U	62	J 392	49678.	-27476.	0.	0.	0.	0.
END	50	U	63	J 394	40697.	0.	-11677.	0.	0.	0.
END	50	D	60	J 391	27896.	0.	0.	0.	0.	0.
END	50	D	61	J 393	-21950.	-14718.	-12980.	0.	0.	0.
END	50	D	62	J 392	8611.	-38729.*	0.	0.	0.	0.
END	50	D	63	J 394	-3145.	0.	-13764.*	0.	0.	0.
MID	NO		60	J 391	37564.	0.	0.	0.	0.	0.
MID	NO		61	J 393	15180.	-12197.	-10689.	0.	0.	0.
MID	NO		62	J 392	37566.	-14316.	0.	0.	0.	0.
MID	NO		63	J 394	15347.	0.	-10713.	0.	0.	0.
1/4	NO		60	J 391	52251.	0.	0.	0.	0.	0.
1/4	NO		61	J 393	2648.	-8271.	-9779.	0.	0.	0.
1/4	NO		62	J 392	26313.	-16447.	0.	0.	0.	0.
1/4	NO		63	J 394	23781.	0.	-9831.	0.	0.	0.
RHE	NO		60	J 391	48388.	0.	0.	0.	0.	0.
RHE	NO		61	J 393	-4579.	-14494.	-11078.	0.	0.	0.
RHE	NO		62	J 392	25433.	-24448.	0.	0.	0.	0.
RHE	NO		63	J 394	15902.	0.	-11523.	0.	0.	0.

\* DENOTES A MINIMUM

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TABLE B59

## SSE TROLLEY REACTIONS

SUM OF ELEMENT FORCES IN ELEMENT CO-ORDINATE SYSTEM (lb, in-lb)

		ELEM	NODE		FX	FY	FZ	MX	MY	NZ
MID	50 U	60 J	391		140210.	0.	0.	0.	0.	0.
MID	50 U	61 J	393		89304.	26111.	21084.	0.	0.	0.
MID	50 U	62 J	392		139030.	26083.	0.	0.	0.	0.
MID	50 U	63 J	394		90170.	0.	21872.	0.	0.	0.
MID	50 D	60 J	391		237360.*	0.	0.	0.	0.	0.
MID	50 D	61 J	393		181970.	31098.	21425.	0.	0.	0.
MID	50 D	62 J	392		236870.	36496.	0.	0.	0.	0.
MID	50 D	63 J	394		182410.	0.	21462.	0.	0.	0.
1/4	50 U	60 J	391		132300.	0.	0.	0.	0.	0.
1/4	50 U	61 J	393		58727.	14893.	17960.	0.	0.	0.
1/4	50 U	62 J	392		92011.	28496.	0.	0.	0.	0.
1/4	50 U	63 J	394		100700.	0.	17631.	0.	0.	0.
1/4	50 D	60 J	391		226460.	0.	0.	0.	0.	0.
1/4	50 D	61 J	393		149140.	21413.	22496.	0.	0.	0.
1/4	50 D	62 J	392		183410.	43023.	0.	0.	0.	0.
1/4	50 D	63 J	394		193040.	0.	22450.	0.	0.	0.
END	50 U	60 J	391		118270.	0.	0.	0.	0.	0.
END	50 U	61 J	393		63596.	28279.	22276.	0.	0.	0.
END	50 U	62 J	392		89566.	38764.	0.	0.	0.	0.
END	50 U	63 J	394		93943.	0.	23158.	0.	0.	0.
END	50 D	60 J	391		197500.	0.	0.	0.	0.	0.
END	50 D	61 J	393		141350.	28468.	24612.	0.	0.	0.
END	50 D	62 J	392		166370.	56766.*	0.	0.	0.	0.
END	50 D	63 J	394		173580.	0.	25603.*	0.	0.	0.
MID	NO	60 J	391		79278.	0.	0.	0.	0.	0.
MID	NO	61 J	393		36341.	16267.	20246.	0.	0.	0.
MID	NO	62 J	392		79320.	18882.	0.	0.	0.	0.
MID	NO	63 J	394		35943.	0.	20280.	0.	0.	0.
1/4	NO	60 J	391		89014.	0.	0.	0.	0.	0.
1/4	NO	61 J	393		18388.	12519.	19507.	0.	0.	0.
1/4	NO	62 J	392		57327.	22714.	0.	0.	0.	0.
1/4	NO	63 J	394		58353.	0.	19705.	0.	0.	0.
RHE	NO	60 J	391		85243.	0.	0.	0.	0.	0.
RHE	NO	61 J	393		38282.	25167.	21264.	0.	0.	0.
RHE	NO	62 J	392		64824.	39439.	0.	0.	0.	0.
RHE	NO	63 J	394		63874.	0.	21903.	0.	0.	0.

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TABLE B60

## SSE TROLLEY REACTIONS

DIFFERENCE OF ELEMENT FORCES IN ELEMENT CO-ORDINATE SYSTEM (lb, in-lb)

	ELEM	NODE		FX	FY	FZ	MX	MY	MZ
MID	50	U	60 J 391	10219.	0.	0.	0.	0.	0.
MID	50	U	61 J 393	4280.	-25891.	-20966.	0.	0.	0.
MID	50	U	62 J 392	11405.	-25864.	0.	0.	0.	0.
MID	50	U	63 J 394	3410.	0.	-21754.	0.	0.	0.
MID	50	D	60 J 391	-86927.	0.	0.	0.	0.	0.
MID	50	D	61 J 393	-88385.	-30879.	-21307.	0.	0.	0.
MID	50	D	62 J 392	-86440.	-36277.	0.	0.	0.	0.
MID	50	D	63 J 394	-88825.*	0.	-21344.	0.	0.	0.
1/4	50	U	60 J 391	44554.	0.	0.	0.	0.	0.
1/4	50	U	61 J 393	8436.	-14554.	-17777.	0.	0.	0.
1/4	50	U	62 J 392	32000.	-28157.	0.	0.	0.	0.
1/4	50	U	63 J 394	19297.	0.	-17448.	0.	0.	0.
1/4	50	D	60 J 391	-49614.	0.	0.	0.	0.	0.
1/4	50	D	61 J 393	-81978.	-21075.	-22313.	0.	0.	0.
1/4	50	D	62 J 392	-59399.	-42684.	0.	0.	0.	0.
1/4	50	D	63 J 394	-73044.	0.	-22267.	0.	0.	0.
END	50	U	60 J 391	55515.	0.	0.	0.	0.	0.
END	50	U	61 J 393	6629.	-27627.	-21924.	0.	0.	0.
END	50	U	62 J 392	37506.	-38113.	0.	0.	0.	0.
END	50	U	63 J 394	22997.	0.	-22806.	0.	0.	0.
END	50	D	60 J 391	-23710.	0.	0.	0.	0.	0.
END	50	D	61 J 393	-71125.	-27817.	-24260.	0.	0.	0.
END	50	D	62 J 392	-39294.	-56115.*	0.	0.	0.	0.
END	50	D	63 J 394	-56641.	0.	-25251.*	0.	0.	0.
MID	NO		60 J 391	21349.	0.	0.	0.	0.	0.
MID	NO		61 J 393	7027.	-16041.	-20124.	0.	0.	0.
MID	NO		62 J 392	21313.	-18656.	0.	0.	0.	0.
MID	NO		63 J 394	7419.	0.	-20158.	0.	0.	0.
1/4	NO		60 J 391	38074.	0.	0.	0.	0.	0.
1/4	NO		61 J 393	-1481.	-12169.	-19317.	0.	0.	0.
1/4	NO		62 J 392	16845.	-22363.	0.	0.	0.	0.
1/4	NO		63 J 394	11471.	0.	-19515.	0.	0.	0.
RHE	NO		60 J 391	37284.	0.	0.	0.	0.	0.
RHE	NO		61 J 393	-16813.	-24256.	-20772.	0.	0.	0.
RHE	NO		62 J 392	13910.	-38528.	0.	0.	0.	0.
RHE	NO		63 J 394	1388.	0.	-21411.	0.	0.	0.

\* DENOTES A MINIMUM



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TABLE B61

OBE ROPE LOADS

SUM OF ELEMENT FORCES IN ELEMENT CO-ORDINATE SYSTEM (lb, in-lb)

		ELEM	NODE		FX	FY	FZ	MX	MY	MZ
MID	50 U	70 J	406		182840.	0.	0.	0.	0.	0.
MID	50 D	70 J	406		367280.*	0.	0.	0.	0.	0.
1/4	50 U	70 J	406		150740.	0.	0.	0.	0.	0.
1/4	50 D	70 J	406		352890.	0.	0.	0.	0.	0.
END	50 U	70 J	406		139200.	0.	0.	0.	0.	0.
END	50 D	70 J	406		333350.	0.	0.	0.	0.	0.
MID	NO	70 J	406		24694.	0.	0.	0.	0.	0.
1/4	NO	70 J	406		23664.	0.	0.	0.	0.	0.
RHE	NO	70 J	406		22256.	0.	0.	0.	0.	0.

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TABLE B62

OBE ROPE LOADS

DIFFERENCE OF ELEMENT FORCES IN ELEMENT CO-ORDINATE SYSTEM (lb, in-lb)

			ELEM	NODE		FX	FY	FZ	MX	MY	MZ
MID	50	U	70	J 406		57192.	0.	0.	0.	0.	0.
MID	50	D	70	J 406		-127250.*	0.	0.	0.	0.	0.
1/4	50	U	70	J 406		89294.	0.	0.	0.	0.	0.
1/4	50	D	70	J 406		-112860.	0.	0.	0.	0.	0.
END	50	U	70	J 406		100840.	0.	0.	0.	0.	0.
END	50	D	70	J 406		-93319.	0.	0.	0.	0.	0.
MID	NO		70	J 406		15307.	0.	0.	0.	0.	0.
1/4	NO		70	J 406		16336.	0.	0.	0.	0.	0.
RHE	NO		70	J 406		17744.	0.	0.	0.	0.	0.

\* DENOTES A MINIMUM

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TABLE 063

SSE ROPE LOADS

SUM OF ELEMENT FORCES IN ELEMENT CO-ORDINATE SYSTEM (lb, in-lb)

	ELEM	NODE		FX	FY	FZ	MX	MY	MZ
MID	50 U	70 J	406	232540.	0.	0.	0.	0.	0.
MID	50 D	70 J	406	583660.*	0.	0.	0.	0.	0.
1/4	50 U	70 J	406	192150.	0.	0.	0.	0.	0.
1/4	50 D	70 J	406	556660.	0.	0.	0.	0.	0.
END	50 U	70 J	406	164930.	0.	0.	0.	0.	0.
END	50 D	70 J	406	519990.	0.	0.	0.	0.	0.
MID	NO	70 J	406	31220.	0.	0.	0.	0.	0.
1/4	NO	70 J	406	28491.	0.	0.	0.	0.	0.
RHE	NO	70 J	406	24622.	0.	0.	0.	0.	0.

\* DENOTES A MAXIMUM



TABLE B64

SSE ROPE LOADS

DIFFERENCE OF ELEMENT FORCES IN ELEMENT CO-ORDINATE SYSTEM (lb, in-lb)

	ELEM	NODE		FX	FY	FZ	MX	MY	MZ
MID	50	U	70 J 406	7496.	0.	0.	0.	0.	0.
MID	50	D	70 J 406	-343630.*	0.	0.	0.	0.	0.
1/4	50	U	70 J 406	47885.	0.	0.	0.	0.	0.
1/4	50	D	70 J 406	-316630.	0.	0.	0.	0.	0.
END	50	U	70 J 406	75100.	0.	0.	0.	0.	0.
END	50	D	70 J 406	-279950.	0.	0.	0.	0.	0.
MID	NO		70 J 406	8781.	0.	0.	0.	0.	0.
1/4	NO		70 J 406	11509.	0.	0.	0.	0.	0.
RHE	NO		70 J 406	15378.	0.	0.	0.	0.	0.

\* DENOTES A MINIMUM





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WHITING REQ. 79604 DATE 8-12-87  
 BY MJM PAGE 8-55 OF 55  
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TABLE B65

ELEMENT LOADS FOR GIRDER A AT THE POINTS  
 OF THE MAXIMUM STRESS WITHIN THE ELEMENT  
 RANGE FROM 22 TO 32 (NOMINAL GIRDER SECTION)

ALL LOADS ARE IN ELEMENT COORDINATE SYSTEM

TROLLEY	LOAD	ELEM	NODE	F <sub>x</sub> (KIP)	F <sub>y</sub> (KIP)	F <sub>z</sub> (KIP)	M <sub>x</sub> (IN KIP)	M <sub>y</sub> (IN KIP)		M <sub>z</sub> (IN KIP)
								SUM	DIFF	
<u>OBE</u>										
MID	UP	28	311	33.8	11.7	11.7	506.7	48199	17811	2230
	DN	28	312	38.4	12.5	13.8	573.4	68131	-2758	2919
1/4	UP	27	310	17.5	2.9	20.0	247.5	29215	18231	1655
	DN	27	310	23.0	3.3	45.3	309.1	47099	347	2268
END	UP	29	312	22.6	7.4	19.4	238.6	16280	11072	1485
	DN	26	309	20.3	3.7	16.9	227.7	26321	2881	1734
<u>SSE</u>										
MID	UP	28	311	51.1	17.1	15.5	690.8	60141	5869	3382
	DN	28	312	58.2	17.8	16.5	797.1	99367	-34393	4148
1/4	UP	27	310	25.3	5.3	26.1	322.2	36862	10584	2177
	DN	27	310	34.4	5.7	71.8	423.6	67607	-20162	3228
END	UP	26	309	32.9	7.2	9.4	369.7	20925	8277	1750
	DN	26	309	35.2	7.5	29.0	369.5	36659	-7456	2512



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WHITING REQ. 79604 DATE 8-12-87  
 BY MJM PAGE B-56 OF 58  
*11/82627*

TABLE B 66

ELEMENT LOADS FOR GIRDER B AT THE POINTS  
 OF THE MAXIMUM STRESS WITHIN THE ELEMENT  
 RANGE FROM 46 TO 56 (NOMINAL GIRDER SECTION)

ALL LOADS ARE IN ELEMENT COORDINATE SYSTEM

TROWEL	LOAD	ELEM	NODE	F <sub>x</sub> (KIP)	F <sub>y</sub> (KIP)	F <sub>z</sub> (KIP)	M <sub>x</sub> (IN KIP)	M <sub>y</sub> (IN KIP)		M <sub>z</sub> (IN KIP)
								SUM	DIFF	
<u>OBE</u>										
MID	UP	52	361	33.6	12.7	15.7	484.0	44941	16764	2707
	DN	52	361	38.0	8.4	14.0	615.7	65456	-3750	3504
1/4	UP	51	360	17.4	5.5	28.3	227.1	29663	18494	1397
	DN	51	360	23.1	6.4	54.4	313.2	47578	580	1926
END	UP	52	362	18.2	9.5	20.0	112.9	15185	10384	2569
	DN	48	358	26.0	16.7	105.0	158.2	27407	3068	16.49
<u>SSE</u>										
MID	UP	54	363	49.0	16.4	118.6	626.2	53600	5151	2827
	DN	52	361	58.2	12.1	17.9	876.0	95760	-34054	4986
1/4	UP	51	360	25.0	9.0	36.7	299.1	37437	10721	1816
	DN	51	360	34.3	10.4	83.0	451.5	68199	-20042	2729
END	UP	52	362	31.7	16.4	23.5	155.6	18500	7070	4700
	DN	52	362	34.6	17.7	44.2	227.9	28763	-3193	4700

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TABLE 867

OBE GIRDER END LOADS

SUM OF ELEMENT FORCES IN ELEMENT CO-ORDINATE SYSTEM (lb. in-lb)

		ELEM	NODE		FX	FY	FZ	MX	MY	MZ	
MID	50	U	18	I	301	36713.	12851.	132940.	573400.	2340600.	2830000.
MID	50	U	35	J	319	31453.	15646.	127770.	703400.	2126700.	3895700.
MID	50	U	42	I	351	33575.	16043.	121620.	625020.	2195600.	3924900.*
MID	50	U	59	J	369	31489.	10918.	116840.	484050.	2250300.	2924400.
MID	50	D	18	I	301	37818.	14942.	183890.	642950.	2711700.	3351700.
MID	50	D	35	J	319	28447.	17463.	178580.	789470.	2350400.	3111300.
MID	50	D	42	I	351	37979.*	19532.	172450.	795200.	2643200.	3871100.
MID	50	D	59	J	369	28622.	14076.	167350.	615700.	2263100.	2927400.
1/4	50	U	18	I	301	22049.	11406.	150770.	510490.	2121600.	1592900.
1/4	50	U	35	J	319	19224.	8595.	69154.	318540.	1394200.	2215500.
1/4	50	U	42	I	351	21956.	20947.	137180.	962820.	2003200.	2130900.
1/4	50	U	59	J	369	19288.	7833.	63996.	227100.	1362300.	1975500.
1/4	50	D	18	I	301	28998.	15587.	234080.	652640.	2863600.	2072700.
1/4	50	D	35	J	319	26584.	11721.	98454.	387440.	1774800.	3030600.
1/4	50	D	42	I	351	29214.	28780.	219210.	1328100.	2759700.	2700000.
1/4	50	D	59	J	369	26659.	10000.	92947.	313200.	1735700.	2725700.
END	50	U	18	I	301	25795.	13018.	157040.	790000.	2375900.	1559200.
END	50	U	35	J	319	22476.	10426.	50488.	321480.	1401200.	2552600.
END	50	U	42	I	351	24872.	29151.	144450.	1434000.	2271500.	2927800.
END	50	U	59	J	369	22368.	15569.	45087.	112940.	1357900.	2774100.
END	50	D	18	I	301	26644.	14022.	241480.*	799980.	2877700.*	1571700.
END	50	D	35	J	319	29950.	14863.	63314.	321060.	1750800.	3678700.
END	50	D	42	I	351	25985.	38544.*	228900.	2007700.*	2760600.	2929200.
END	50	D	59	J	369	29990.	15788.	57879.	158150.	1700900.	3221900.
MID	NO		18	I	301	25105.	8493.	82193.	416430.	1609500.	1897300.
MID	NO		35	J	319	16318.	9921.	77726.	499360.	1323500.	1792800.
MID	NO		42	I	351	25368.	11066.	72515.	450750.	1537700.	2211000.
MID	NO		59	J	369	16409.	7974.	67969.	349010.	1272400.	1675800.
1/4	NO		18	I	301	22060.	8899.	101990.	431420.	1627700.	1326000.
1/4	NO		35	J	319	14876.	6669.	52031.	277320.	1082300.	1738900.
1/4	NO		42	I	351	21534.	16398.	89193.	747530.	1534200.	1824400.
1/4	NO		59	J	369	14925.	6594.	46985.	176310.	1048700.	1533900.
RHE	NO		18	I	301	24589.	14574.	107570.	810510.	1986900.	1419400.
RHE	NO		35	J	319	28387.	10172.	40150.	295220.	1575600.	2596800.
RHE	NO		42	I	351	25813.	32938.	96593.	1349200.	1821400.	2950100.
RHE	NO		59	J	369	28053.	19222.	34926.	45552.	1520400.	3725200.

\* DENOTES A MAXIMUM

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TABLE 8 68

## SSE GIRDER END LOADS

SUM OF ELEMENT FORCES IN ELEMENT CO-ORDINATE SYSTEM (lb, in-lb)

			ELEM	NODE		FX	FY	FZ	MX	MY	MZ
MID	50	U	18	I	301	57798.	17955.	164000.	762110.	3078800.	3883900.
MID	50	U	35	J	319	49060.	22282.	157680.	946480.	2805700.	6291300.
MID	50	U	42	I	351	51328.	21542.	149840.	808370.	2836700.	5933300.
MID	50	U	59	J	369	49034.	13950.	144280.	626190.	3107000.	4388700.
MID	50	D	18	I	301	57835.	21290.	263230.	875340.	3807500.	4775100.
MID	50	D	35	J	319	40584.	24870.	256530.	1083900.	3237100.	4450400.
MID	50	D	42	I	351	58175.*	27797.	248510.	1131300.	3724800.	5516700.
MID	50	D	59	J	369	40827.	20028.	242200.	875960.	3102700.	4166900.
1/4	50	U	18	I	301	36850.	14991.	185460.	685390.	2793500.	2260500.
1/4	50	U	35	J	319	25017.	11540.	85295.	420390.	1673800.	2942500.
1/4	50	U	42	I	351	35935.	27566.	168110.	1268100.	2652100.	3138100.
1/4	50	U	59	J	369	25114.	11503.	78998.	299140.	1635100.	2579000.
1/4	50	D	18	I	301	48302.	22169.	329250.	935280.	4110500.	3048400.
1/4	50	D	35	J	319	37878.	16861.	134610.	541780.	2367400.	4345200.
1/4	50	D	42	I	351	47440.	41197.	310980.	1914700.	4021500.	4054600.
1/4	50	D	59	J	369	37989.	15089.	128040.	451530.	2309300.	3891300.
END	50	U	18	I	301	48868.	22824.	180610.	1392200.	3321500.	2832600.
END	50	U	35	J	319	32414.	13751.	60077.	507080.	1842800.	3228500.
END	50	U	42	I	351	46736.	43470.	164810.	1976400.	3224000.	5372000.
END	50	U	59	J	369	31987.	28267.	53522.	155580.	1784800.	4616300.
END	50	D	18	I	301	49965.	24123.	330690.*	1403800.	4213600.†	2847500.
END	50	D	35	J	319	44051.	21069.	81362.	504230.	2402000.	5135900.
END	50	D	42	I	351	48101.	57701.†	316400.	2894200.‡	4097300.	5373400.
END	50	D	59	J	369	43930.	28531.	75158.	227920.	2326500.	5234300.
MID	NO		18	I	301	39697.	11090.	101010.	520220.	2093500.	2471100.
MID	NO		35	J	319	21482.	12930.	95709.	628360.	1578600.	2382100.
MID	NO		42	I	351	40332.	14376.	88631.	585300.	1993900.	2895800.
MID	NO		59	J	369	21585.	10353.	83137.	453190.	1525100.	2189900.
1/4	NO		18	I	301	38721.	12007.	120310.	590990.	2186700.	1971500.
1/4	NO		35	J	319	19833.	9209.	61845.	366440.	1315200.	2387100.
1/4	NO		42	I	351	37142.	22264.	104210.	1010800.	2109000.	2819200.
1/4	NO		59	J	369	19910.	10117.	55824.	238410.	1271200.	2055000.
RHE	NO		18	I	301	45144.	25976.	118580.	1278000.	2852300.	2519000.
RHE	NO		35	J	319	49500.	15167.	47411.	430290.	2504700.	4078000.
RHE	NO		42	I	351	47713.	57245.	105550.	2098400.	2633100.	5172300.
RHE	NO		59	J	369	48769.	34657.	41236.	70761.	2431000.	6697100.*

\* DENOTES A MAXIMUM





WHITING REQ. 79604 DATE 8-21-87  
 BY MJM PAGE C-1 OF 3  
*8/24/87*

APPENDIX 'C'

NOMENCLATURE & REFERENCES

NOMENCLAUTRE

A	=	Area
a	=	Length of section that is buckling
b	=	Length of section that is buckling
c	=	Distance to extreme fibers
d	=	Distance to N.A.
E	=	Modulus of elasticity
F <sub>x</sub>	=	Force applied in the "x" direction
F <sub>y</sub>	=	Force applied in the "y" direction
F <sub>z</sub>	=	Force applied in the "z" direction
I <sub>x</sub>	=	Moment of Inertia about "x-x" axis
I <sub>y</sub>	=	Moment of Inertia about "y-y" axis
I <sub>z</sub>	=	Moment of Inertia about "z-z" axis
J <sub>x</sub>	=	Polar Moment of Inertia about "x-x" axis
J <sub>y</sub>	=	Polar Moment of Inertia about "y-y" axis
J <sub>z</sub>	=	Polar Moment of Inertia about "z-z" axis
L	=	Length
M <sub>x</sub>	=	Moment about "x" axis
M <sub>y</sub>	=	Moment about "y" axis
M <sub>z</sub>	=	Moment about "z" axis
P	=	Load
ROTX	=	Rotation about the Global "x" axis used in modal analysis
ROTY	=	Rotation about the Global "y" axis used in modal analysis
ROTZ	=	Rotation about the Global "z" axis used in modal analysis
UX	=	Displacement in Global "x" direction used in modal analysis
UY	=	Displacement in Global "y" direction used in modal analysis
UZ	=	Displacement in Global "z" direction used in modal analysis
$\bar{x}$	=	Centroid location in "x" direction
$\bar{y}$	=	Centroid location in "y" direction
$\bar{z}$	=	Centroid location in "z" direction
$\sigma$	=	Tensile or Compressive Stress
$\tau$	=	Shear Stress

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WHITING REQ. 79604 DATE 8-21-87  
BY MJM PAGE C-2 OF 3  
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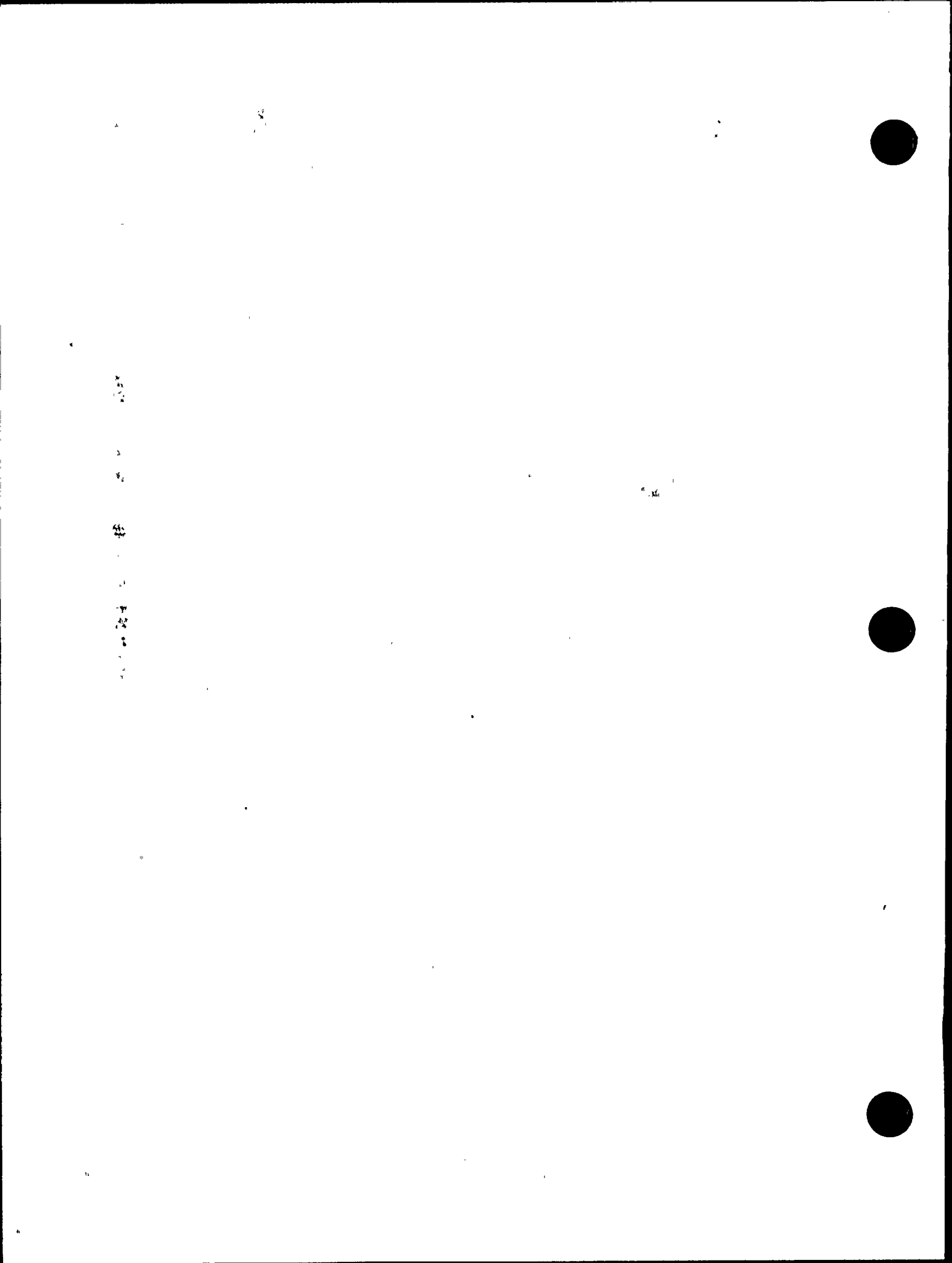
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WHITING REQ. 79604 DATE 8-21-87  
 BY MJM PAGE C-3-R1 OF 3  
*RMH 8-24-87*  
 REV MJM 9-15-87 *RMH 9-15-87*

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WHITING REQN. 79604 DATE 8-26-87  
BY MJM PAGE D-1 OF 13  
*CA 8.2.87*

APPENDIX D  
EVALUATION OF 55 TON LOAD

After the completion of all computer runs with a 50 ton load, it was requested that the crane be evaluated with a 55 ton load during a seismic event. The initial runs indicated that the SSE produces stresses that are closer to the allowables than the OBE. The trolley at midspan with the load down produces the maximum girder stresses and the trolley at end with the load down produces the maximum wheel loads. Therefore these two cases were run with a 55 ton load but with all other items as in the body of the report for the 50 ton load.

The mode coefficients for these runs are shown in tables D1 and D2 and the unfactored and factored reactions are shown in tables D3 through D6. The wheel loads were derived as described in the body of the report and tabulated in table D7. The maximum stresses from the computer runs are summarized in tables D8 and D9. The rope loads are summarized in table D10.

The results for trolley at mid are summarized and compared with the results for 50T and with the allowables in table D11. The results for trolley at end are summarized and compared with the results for 50T and with the allowables in table D12. For those components which were analyzed manually with loadings from the computer program the highest stresses were proportioned from the analysis for 50T by applying a multiplier for the appropriate component from tables D11 and 12. The results are shown in table D13. The girder web stability was similarly proportioned as shown in table D14.

It was found that the stresses in the crane did not exceed the allowable stresses when loaded with a 55 ton load during the specified seismic event.



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TABLE D1

SUMMARY OF NATURAL FREQUENCIES AND MODE COEFFICIENTS - PAOMNDS, 55

MODE FREQUENCY MODE COEFFICIENT FOR SPECIFIED DIRECTION  
 HZ X Y Z

1	2.02	0.0955 *	241.0000 * MAX	0.2006 *
2	2.83	3.4330 * MAX	0.1574 *	88.8900 * MAX
3	4.14	1.3270 *	1.0900 *	0.0461 *
4	6.32	0.3182 *	0.0343	1.5750 *
5	7.99	0.1032 *	0.1987 *	0.0602 *
6	9.05	0.0441 *	0.0480	0.0054
7	9.64	0.0378 *	0.0017	0.0225
8	11.59	0.0325 *	0.0482	0.0007
9	13.31	0.0403 *	0.0135	0.0024
10	15.09	0.2849 *	0.0011	0.0129
11	18.04	0.0054	0.0177	0.0020
12	23.32	0.0012	0.0104	0.0006
13	28.72	0.0005	0.0011	0.0000
14	31.23	0.0077	0.0003	0.0015
15	35.24	0.0012	0.0003	0.0002
16	43.69	0.0002	0.0001	0.0003
17	54.52	0.0005	0.0003	0.0001
18	56.18	0.0005	0.0001	0.0002
19	61.02	0.0006	0.0000	0.0004
20	68.89	0.0005	0.0001	0.0008
21	75.33	0.0002	0.0000	0.0002
22	83.28	0.0002	0.0001	0.0001
23	85.97	0.0002	0.0001	0.0002
24	90.22	0.0001	0.0000	0.0001
25	91.85	0.0000	0.0000	0.0001
26	91.91	0.0000	0.0000	0.0003
27	123.90	0.0000	0.0000	0.0000
28	174.80	0.0000	0.0000	0.0000

SIGNIFICANCE FACTOR 0.50% 0.05% 0.05%

\* INDICATES EXPANDED MODE

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY MID, 55T LD DN / SSE 'X'

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	1294.	50169.	551.	50189.	0.	50189.	-50189.
101 FZ	8341.	6691.	174882.	175208.	104525.	279734.	-70683.
101 MX	114225.	3351455.	130566.	3355941.	44352.	3400293.	-3311589.
102 FZ	7915.	6645.	168438.	168755.	96374.	265128.	-72381.
102 MX	93993.	153495.	20605.	181163.	90341.	271504.	-90822.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	61612.	1563.	18358.	64308.	0.	64308.	-64308.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	1694.	45891.	862.	45931.	-0.	45931.	-45931.
201 FZ	7356.	5835.	172091.	172347.	100800.	273147.	-71547.
201 MX	76239.	3097556.	103406.	3100220.	43474.	3143693.	-3056746.
202 FZ	6936.	5874.	165998.	166246.	92648.	258895.	-73598.
202 MX	50304.	67267.	32586.	90095.	89459.	179555.	-636.

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY MID, 55T LD DN / SSE 'X'

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
NODE LABEL							
101 FY	1294.	453119.	551.	453121.	0.	453121.	-453121.
101 FZ	8341.	59719.	174882.	184985.	104525.	289511.	-80460.
101 MX	114225.	30263769.	130566.	30264266.	44352.	30308618.	-30219914.
102 FZ	7915.	58902.	168438.	178615.	96374.	274989.	-82242.
102 MX	93993.	1280460.	20605.	1284071.	90341.	1374412.	-1193730.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	61612.	1600.	18358.	64309.	0.	64309.	-64309.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	1694.	414417.	862.	414421.	-0.	414421.	-414421.
201 FZ	7356.	51905.	172091.	179899.	100800.	280699.	-79098.
201 MX	76239.	27976096.	103406.	27976391.	43474.	28019864.	-27932917.
202 FZ	6936.	51651.	165998.	173986.	92648.	266634.	-81337.
202 MX	50304.	408174.	32586.	412551.	89459.	502011.	-323092.

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79604/MJM/AEP, DCC, OLD MAIN TROLLEY END, 55T LD DN / SSE 'X'

REACTION SUMMARY

LOAD STEP	1	2	3	SRSS	STATIC	SUM	DIFFER
101 FY	18235.	76898.	3286.	79099.	0.	79099.	-79099.
101 FZ	22850.	7448.	201904.	203330.	152746.	356076.	-50583.
101 MX	1104536.	5161683.	262544.	5285064.	41780.	5326844.	-5243283.
102 FZ	21852.	8569.	198536.	199919.	141556.	341475.	-58363.
102 MX	167002.	1479978.	26226.	1489602.	91070.	1580672.	-1398532.
123 FY	0.	0.	0.	0.	0.	0.	0.
123 FZ	0.	0.	0.	0.	0.	0.	0.
123 MX	0.	0.	0.	0.	0.	0.	0.
123 MY	0.	0.	0.	0.	0.	0.	0.
124 FX	47259.	7482.	34951.	59253.	0.	59253.	-59253.
124 FY	0.	0.	0.	0.	0.	0.	0.
124 FZ	0.	0.	0.	0.	0.	0.	0.
124 MX	0.	0.	0.	0.	0.	0.	0.
124 MY	0.	0.	0.	0.	0.	0.	0.
124 MZ	0.	0.	0.	0.	0.	0.	0.
201 FY	8098.	37335.	1018.	38217.	-0.	38217.	-38217.
201 FZ	9986.	3785.	38378.	39836.	52632.	92469.	12796.
201 MX	283481.	2425952.	44822.	2442870.	37973.	2480843.	-2404897.
202 FZ	9086.	3811.	37109.	38395.	47413.	85808.	9018.
202 MX	48462.	1290425.	24622.	1291569.	85057.	1376626.	-1206512.



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WHITING REQ. 79604 DATE 8-26-87  
 BY MJM PAGE D-8 OF 13  
AS2 8-27-87

TABLE D-7

CRANE WHEEL LOADS  
 55 T LOAD

SSE SCALED

	TROLLEY	LOAD	W <sub>X</sub> MAX	W <sub>Y</sub> MAX	MAX W <sub>Z</sub>		P <sub>UL</sub> 201,202	P <sub>UR</sub> 101,102	TABLE USED
					W <sub>A</sub> (MAX)	W <sub>B</sub> **			
DRIVER 101,201	MID	DN	10.7	49.1	196.5	83.2	32.8	35.7	D4
	END	DN	9.9	66.7	266.8	89.3	19.4	37.7	D6
IDLER 102,202	MID	DN	10.7	-	138.2	126.9	-	-	D4
	END	DN	9.9	-	203.7	137.8	10.4	-	D6

ALL FORCES IN KIPS IN GLOBAL COORDINATE SYSTEM

\* INDICATES UPKICK LOAD AT UPKICK LUG FOR STATIC PLUS DYNAMIC

\*\* W<sub>B</sub> IS LOAD ON OTHER WHEEL OF TRUCK WHEN W<sub>A</sub> IS MAX

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TABLE D8

MAXIMUM STRESSES FROM PFAOMMDS 55

SSE MID 55 D

COMPONENT	ELEM	NODE	X	Y	Z	SRSS	STATIC	SUM
GIRDER A	28	312	1049.	7590.	20983.	22338.	10132.	32470.
GIRDER B	52	361	985.	9096.	20341.	22304.	9609.	31913.
END CONNECT-RHE	17	154	1661.	18032.	345.	18111.	506.	18617.
END CONNECT-LHE	74	252	1905.	15340.	480.	15465.	305.	15770.

TABLE D9

MAXIMUM STRESSES FROM PEAOEMDS 55

SSE END 55 D

COMPONENT	ELEM	NODE	X	Y	Z	SRSS	STATIC	SUM
GIRDER A	35	319	4172.	13583.	665.	14225.	181.	14406.
GIRDER B	43	353	1796.	12846.	4008.	13576.	2359.	15935.
END CONNECT-RHE	17	154	1001.	18685.	465.	18718.	471.	19188.
END CONNECT-LHE	74	252	5316.	17455.	668.	18258.	358.	18616.

STRESS IN PSI

ALLOWABLE STRESS = 36000 = 32700 PSI



WHITING REQN. 79604 DATE 8-26-87  
 BY MJM PAGE D-10 OF 13  
 ASZ 8-27-87

TABLE D-10

ROPE LOADS  
 55 T LOAD

SSE SCALED

TROLLEY	LOAD	STATIC + DYNAMIC KIP	STATIC - DYNAMIC KIP
MID	DN	623.2	-363.2
END	DN	559.2	-299.2

MAX

$$623.2 \text{ KIP} = 311.6 \text{ T}$$

ALLOWABLE 1053 T

$$R = \frac{311.6}{1053} = .30$$

$$DLF = \frac{311.6}{65} = 4.79$$

$$DLF' = \frac{311.6}{160} = 1.95$$

WHITING REQ. 79604 DATE 3-27-37  
 BY MJM PAGE D-11 OF 13  
~~8-27-87~~

TABLE D-11

COMPARISON BETWEEN 50 AND 55 T - SSE  
 TROLLEY AT MID LOAD DOWN

COMPONENT	ITEM	50 T		55 T		RATIO 55T 50T	ALLOW	RATIO STRESS ALLOW	
		LOAD STR	REF TABLE	LOAD STR	REF TABLE				
WHEEL DR	WX MAX	10.8	4-5	10.7	D7	.99	-	-	
	WY MAX	47.8	4-5	49.1	D7	1.03	-	-	
	WZ MAX	191.2	4-5	196.5	D7	1.03	-	-	
	PO MAX	35.4	4-5	35.7	D7	1.01	-	-	
	IO	WX MAX	10.8	4-5	10.7	D7	.99	-	-
		WY MAX	-	4-5	-	D7	-	-	-
		WZ MAX	134.1	4-5	138.2	D7	1.03	-	-
		PO MAX	-	4-5	-	D7	-	-	-
GIRDER A	σ	31.5	B5	32.5	D8	1.03	32.7	.99	
	B	30.9	B5	31.9	D8	1.03	32.7	.98	
END CON R	σ	18.1	B5	18.6	D8	1.03	32.7	.57	
	L	15.4	B5	15.8	D8	1.03	32.7	.48	
ROPE SUM	P MAX	583.7	B63	623.2	D10	1.07	1053x2	.30	
	DIFF	P MIN	-343.6	B64	-363.2	D10	1.06	-	-

LOADS IN KIPS  
 STRESS IN KSI





WHITING REQ. 79604 DATE 3-27-97  
 BY MJM PAGE D-12 OF 13  
~~1/27/827-87~~

TABLE D-12

COMPARISON BETWEEN 50 AND 55 T - SSE  
 TROLLEY AT END LOAD DOWN

COMPONENT	ITEM	50 T		55 T		RATIO 55T 50T	ALLOW	RATIO STRESS ALLOW	
		LDOR STR	REF TABLE	LDOR STR	REF TABLE				
WHEEL DR	WX MAX	9.9	4-5	9.9	D7	1.00	-	-	
	WY MAX	64.2	4-5	66.7	D7	1.04	-	-	
	WZ MAX	256.8	4-5	266.8	D7	1.04	-	-	
	P <sub>U</sub> MAX	36.6	4-5	37.7	D7	1.03	-	-	
	ID	WX MAX	9.9	4-5	9.9	D7	1.00	-	-
		WY MAX	-	4-5	-	D7	-	-	-
		WZ MAX	195.2	4-5	203.7	D7	1.04	-	-
		P <sub>U</sub> MAX	10.6	4-5	10.4	D7	.98	-	-
GIRDER	A	5	817	14.4	D9	1.04	32.7	.44	
	B	5	817	15.9	D9	1.01	32.7	.49	
END CON	R	5	817	19.2	D9	1.00	32.7	.59	
	L	5	817	18.6	D9	1.01	32.7	.57	
ROPE SUM	P <sub>MAX</sub>	520.0	863	559.2	D10	1.08	1053x2	.27	
	P <sub>U</sub>	-280.0	864	-299.2	D10	1.07	-	-	

LOADS IN KIPS  
 STRESS IN KSI

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WHITING REQN. 79604 DATE 8-27-87  
 BY MJM PAGE D13 OF 13  
ASZ 8-27-87

TABLE D13

RATIOED STRESSES  
 FOR COMPONENTS ANALYZED MANUALLY  
 FROM COMPUTER LOADINGS

SSE

COMPONENT	DETAIL	REF PAGE	STRESS 50 T	MULT	STRESS 55 T	ALLOW	$\frac{\text{STRESS}}{\text{ALLOW}}$
BR AXLE	SHEAR	4-15	9.9	1.04	10.3	32.7	.31
SEISMIC LUG	R TENS	4-19	15.1	1.03	15.6	32.7	.48
BR TRUCK	TENS	4-33	16.8	1.04	17.5	32.7	.54
TAK TO GIRD	BOLT	4-52	46.2	1.04	48.0	50.2	.96
"	WELD	4-70	13.2	1.04	13.7	27.3	.50
GIRD TO ET	BOLT	4-84	47.7	1.03	49.1	50.2	.98
"	WELD	4-92	23.2	1.03	23.9	27.3	.88
GIRDER END	SHEAR	4-105	13.8	1.04	14.4	19.6	.73

STRESS IN KSI

TABLE D.14

RATIOED GIRDER BUCKLING STABILITY

SSE

COMPONENT	DETAIL	REF PAGE	STABILITY RATIO 50 T	MULT	STABILITY RATIO 55 T	ALLOW	$\frac{\text{RATIO}}{\text{ALLOW}}$
GIRDER WEB	PANEL 2	4-103	.876	1.03	.902	.909	.99

