



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|-------------------|----------------|--------------------------------|--|---|--|---|---|---------------|
| 10029091 | (ATL) AVIATION TRADERS LIMITED | P-EASA.A.S.03891 | | | B747-400 | Introduction of changes to the cabin layout in accordance with ATL Modification Summary MOD-11427-00001 Part A: introduces a seating configuration of 22 business class and 44 economy class Part B: introduces all emergency equipment necessary to comply (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The certification basis is CS25 at change 4 for changed parts except for the following re (...) | EASA AFM supplement 11427-AFM-0001 Issue 1, Installation of Photo-luminescent Aisle Path Marking System, shall be used with the aircraft instructions for Continued Airworthiness included in ATL Mod Summary MOD11427-00001 Issue 4 shall be incorporated in (...) | None | 03.03.2010 |
| 10029662 | (ATL) AVIATION TRADERS LIMITED | P-EASA.A.S.03876 | | | B737-300 | Business Interior - 48 PAX LOPA | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-01/Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane Systems" (...) | Modification Summary Sheet: MOD-11348-00001 Issue 2 dated 30 March 2010 - Master Document List MDL-11348-00001 Issue 2 dated 29 March 2010 - Airplane Flight Manual Supplement 11348-AFM-00001 issue 2, or later revisions of the above listed documents approved by EASA | The STC holder shall provide by 7th June 2010 Instructions for Continued Airworthiness on Electrical Wiring Interconnection System - EWIS ICA - in accordance with Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane Systems" - ICA on E (...) | 09.04.2010 |
| 10029965 | (ATL) AVIATION TRADERS LIMITED | | | | B757-200 | Boeing 757-200 229Y Passenger Configuration | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-01/Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane Systems" - ICA on EWIS | ATL Modification Summary Sheet, Doc. No. MOD-11682-00001, Issue 3 dated 6th May 2010 - ATL Master Document List, Doc. No. MDL-11682-00001, Issue 3 dated 5th May 2010 or later revisions of the above listed documents approved by EASA | As per associated Technical documentation for weight and balance (maximum allowable limit in the forward and aft cargo area is to be amended as per indicated in doc. N° MOD-11682-00001, Issue 3). Limited to: MSN 27973 | 07.05.2010 |
| 10031637 | 1263343 ALBERTA INC DBA ENERJET | | | TCCA STC C-LSA10-077/D ISSUE 1 | B737-700 | Bucher LH Windscreen /Closet Modification | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-01 Issue 1: Special Condition - EWIS ICA The requirements for environmental protection (...) | Modification Data Summary R1004821 issue A dated April 2010 Modification Review Continued Airworthiness & Safety Improvement R1006846 issue A dated June 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED De (...) | Limited to B737-700 aircraft S/N 33920 Maintenance of the Bucher Closet is to be performed by Boeing Aircraft/Maintenance Manual D633A101-CQT <-> <-> and Bucher Component Maintenance Manual 25-30-65 | 01.09.2010 |
| 10031638 | 1263343 ALBERTA INC DBA ENERJET | | | TCCA STC C-LSA08-311/D ISSUE | B737-700 | 131, 137 <-> <-> 145 Passenger Seating Reconfiguration | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-01 Issue 1: Special Condition - EWIS ICA The requirements for environmental protection (...) | Modification Data Summary R0810701 issue C dated April 2010 Modification Review Continued Airworthiness <-> <-> Safety Improvement R1006845 issue A dated June 2010 or later revisions of the above listed documents approved by EASA in accordance with EAS (...) | Limited to B737-700 aircraft S/N 33920 Maintenance of the Racaro seats are to be performed per Racaro Component Maintenance Manual 25-22-11, 25-25-81 and 25-24-79 | 01.09.2010 |
| 10031639 | 1263343 ALBERTA INC DBA ENERJET | | | TCCA STC C-LSA08-310/D ISSUE | B737-700 | Artes C406-2 ELT Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-01 Issue 1: Special Condition - EWIS ICA The requirements for environmental protection (...) | Modification Data Summary R0810709 issue A dated November 2008 - Modification Review Continued Airworthiness & Safety Improvement R1006844 issue A dated June 2010 - Aircraft Maintenance Manual Supplement R0809001 issue A - ARTEX Maintenance Manual 570- (...) | Limited to B737-700 aircraft S/N 33920 and 34480 | 02.09.2010 |
| 10031640 | 1263343 ALBERTA INC DBA ENERJET | | | TCCA STC C-LSA10-078/D ISSUE | B737-700 | TIA Model 1545 oven installation in forward LH monument. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-01 Issue 1: Special Condition - EWIS ICA The requirements for environmental protection (...) | Modification Data Summary R1004825 issue A dated April 2010 Modification Review Continued Airworthiness <-> <-> Safety Improvement R1006847 issue A dated June 2010 or later revisions of the above listed documents approved by EASA in accordance with EAS (...) | Limited to B737-700 aircraft S/N 33920 Maintenance of the TIA Component Oven is to be performed per TIA Component Maintenance Manual 25-30-61 | 01.09.2010 |
| 10014074 | 328 DESIGN GMBH | EASA.A.S.01166 | REV. 1 | | BAE JETSTREAM 31 BAE JETSTREAM 32 | Installation of Airborne Collision Avoidance System (ACAS II). This certificate follows EASA.A.S.01166 due to a transfer of ownership. | | -Avionics Mobile Services Definition Document Part 21-002/1485-01 Issue 1 or later approved revision. Avionics Mobile Services AFM Supplement Part 21-008/011 Issue 1 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.03.2013 |
| 10014139 | 328 DESIGN GMBH | EASA.A.S.01252 | REV. 1 | | CESSNA 550 CESSNA 650 | Installation of an Airborne Collision Avoidance System (ACAS II). This certificate follows EASA.A.S.01252 due to a transfer of ownership. | | Avionics Mobile Services Project Certification Plan AM/1493. Avionics Mobile Services AFM Supplement Number Part21-008/008. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.03.2013 |
| 10014184 | 328 DESIGN GMBH | EASA.IM.A.S.01322 | REV. 1 | | BAE 125-800 | Installation of an Airborne Collision Avoidance System (ACAS II). This certificate follows EASA.IM.A.S.01322 due to a transfer of ownership. | | Avionics Mobile Services Project Certification Plan AM/1498 Reference PART21-003/1498-01 or later approved revision. Avionics Mobile Services AFM Supplement reference AFM PART21-008 Issue 2 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.03.2013 |
| 10014198 | 328 DESIGN GMBH | EASA.A.S.01341 | REV. 1 | | BAE 125-800 | Installation of an Airborne Collision Avoidance System (ACAS II). This certificate follows EASA.A.S.01341 due to a transfer of ownership. | | Avionics Mobile Services Project Certification Plan AM/1498 Reference PART21-003/1498-01 or later approved revision. Avionics Mobile Services ARM Supplement reference ARM PART21-008 Issue 2 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.03.2013 |
| 10014215 | 328 DESIGN GMBH | EASA.A.S.01376 | REV. 1 | | CESSNA 550 SERIES CESSNA 560 SERIES | Avionics Mobile Services Modification AM/1478. Installation of an Airborne Collision Avoidance System (ACAS II). This certificate follows EASA.A.S.01376 due to a transfer of ownership. | | -Avionics Mobile Services Ltd Project Certification Plan AM/1478 reference PART21-003/1478-01-Avionics Mobile Services Ltd AFM Supplement PART21-008/008 dated April 08 2005 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.03.2013 |



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| 10014303 | 328 DESIGN GMBH | EASA.A.S.01547 | REV. 1 | | PIPER PA-31T2 | Certification of FAA Approved field modifications for the issue of anEASA C of A,Avionics Mobile Services AFM supplement Part 21-008 014 GNS-530,Avionics Mobile Services AFM supplement Part 21-008 015 KMH-880,Avionics Mobile Services AFM supplement (...) | | This STC is approved only for the product configuration as defined inthe approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall bedetermined by the installer. | This STC is approved only for the product configuration as defined inthe approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall bedetermined by the installer. | 11.03.2013 |
| 10014431 | 328 DESIGN GMBH | EASA.A.S.02479 | REV. 1 | | 1124A WESTWIND II | Installation of Airborne Collision Avoidance System (ACAS II).This certificate follows EASA.A.S.02479 due to a transfer of ownership. | | Avionics Mobile Services Modification AM/1486,Avionics Mobile Services AFM Supplement No. PART21-008/017 Revision - dated April 08 2005 or later approved revision. | This STC is approved only for the product configuration as defined inthe approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall bedetermined by the installer. | 11.03.2013 |
| 10014480 | 328 DESIGN GMBH | EASA.A.S.02543 | REV. 3 | | HS 125-700 SERIES | Installation of an Airborne Collision Avoidance System (ACASII).This certificate follows EASA.A.S.02543 due to a transfer of ownership. | | Avionics Mobile Services Limited Modification AM/1517,Avionics Mobile Services Limited AFM Supplements ref. PART21-008/018Issue 1 or later approved revision & PART21-008/001 Issue 1 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 11.03.2013 |
| 10014481 | 328 DESIGN GMBH | EASA.A.S.02544 | REV. 1 | | HS 125-700 SERIES | Installation of a Digital Air Data Display System.This certificate follows EASA.A.S.02544 due to a transfer of ownership. | | Avionics Mobile Services Modification AM/1518,Avionics Mobile Services AFM Supplement PART21-008/019 Issue 1 or later approved revision | This STC is approved only for the product configuration as defined inthe approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall bedetermined by the installer. | 11.03.2013 |
| 10015249 | 328 DESIGN GMBH | EASA.A.S.03688 | REV. 1 | | BAE JETSTREAM 41 | Mode-S Enhanced Surveillance (EHS) Upgrade | | Certification Program AM1577/003A-01Instructions for Continued Airworthiness AM1557/007A-01Flight Manual Supplement AM1557/008A-01 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 15.04.2013 |
| 10015654 | 328 DESIGN GMBH | EASA.IM.A.S.0163 | REV. 2 | | HS 125-700 SERIES | Installation of a Rockwell Collins TCAS-4000 Traffic Alert and CollisionAvoidance System (TCAS-II)This certificate follows EASA.IM.A.S.0163 due to a transfer of ownership. | | Avionics Mobile Services Limited modification AM/1469,Avionics Mobile Services Limited AFM Supplement, reference:PART21-008/001 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 11.03.2013 |
| 10015663 | 328 DESIGN GMBH | EASA.IM.A.S.0164 | REV. 2 | | HS 125-700 SERIES | Installation of a Ground Proximity Warning System (GPWS) and TerrainAwareness Warning System (TAWS).This certificate follows EASA.IM.A.S.0164 due to a transfer of ownership. | | Avionics Mobile Services Limited modification AM/1470,Avionics Mobile Services Limited AFM Supplement, reference: PART21-008/006 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 11.03.2013 |
| 10031570 | 328 DESIGN GMBH | P-EASA.A.S.03968 | REV. 1 | | DORNIER 328-100, 328-300 | Installation of Emergency Lighting Power Supply System | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CRI H-01/SpecialCondition H-01 "Enhanced Airworthiness Programme forAeroplane Systems (...) | Master Document List - AM1561010A-01 Issue 2 dated 16 AUG 2010AFM Supplement - AM1561008A-01 Issue 1Instructions for Continued Airworthiness - AM1561007A Issue Aor later revisionsof the above listed documents approved by EASA | None | 15.04.2013 |
| 10036511 | 328 DESIGN GMBH | | REV. 1 | | HAWKER 800XP | Installation of a Flight Data Recorder (FDR) System. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Installation in accordance with Avionics Mobile Design ServicesEngineering Order ref AM1581009A-01 Issue A dated 05 August 2011.Operation in accordance with Avionics Mobile Design Services AircraftFlight Manual Supplement ref AM1581008A-01, dated, Ju (...) | None. | 15.04.2013 |
| 10043118 | 328 DESIGN GMBH | EASA.A.S.01322 | REV. 1 | | CN-235M-100 SERIES | Installation of a Ground Proximity Warning System (GPWS) & TerrainAwareness Warning System (TAWS).This certificate follows EASA.A.S.01322 due to a transfer of ownership. | | Avionics Mobile Services Definition Document PART21-002/1504-01, Issue1, dated September 9, 2005 or later approved revision,Avionics Mobile Services AFM Supplement PART21-008-012,Revision 1 orlater approved revision. | 1. Aircraft Registrations A40-CU and A40-CV.2. This STC is approved only for the product configuration as defined intheapproved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall bedetermined by the installer. | 11.03.2013 |
| 10043137 | 328 DESIGN GMBH | 2003-2150 | REV. 1 | | BAE JETSTREAM 31 | Installation of Honeywell MK VI Enhanced Ground Proximity WarningSystem. STC CAA.21NE2.00044.This STC follows the EASA approval letter 2150 of December 2003 due to atransfer of ownership. | | | | 11.03.2013 |
| 10043139 | 328 DESIGN GMBH | 2004-2293 | REV. 1 | | BAE 125-800B | Installation of a Honeywell Airborne Collision Avoidance System (ACASII).This STC follows EASA approval letter 2004-2293 due to a transfer of ownership. | | | | 11.03.2013 |
| 10043140 | 328 DESIGN GMBH | 2004-5548 | REV. 1 | | DORNIER 328-100 | Installation of a Honeywell MK VI Enhanced Ground ProximityWarningSystem (EGPWS).This STC follows EASA approval letter2004-5548 due to a transfer of ownership. | | STC Definition Document PART21-003/1459-01. | | 11.03.2013 |



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| 10044141 | 328 DESIGN GMBH | | | | CL-600-2B19 | Cabin Modification to VIP. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:- the following paragraph(s) at a later amendment CS 25.1701 to CS 25.1729 at Amendment 11. The requirements for environme (...) | Declaration of Compliance, Doc. No. DC-10D003Z0001, Issue Bdated20.03.2013, Master Document List, Doc. No. MDL003Z0001, Issue A dated 19.03.2013, Instructions for Continued Airworthiness, Doc. No. ICA010D259Z0001, Issue A dated 19.03.2013, AFM Supplement (...) | Aircraft Serial Number 7617 only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 20.03.2013 |
| 10044717 | 328 DESIGN GMBH | | REV.1. | | GIV | Cabin Refurbishment with LOPA Change, Cabin Lighting, Cabin Power Supply System, and new In-Flight Entertainment. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with 328DO Engineering Order EO-16H259Z0001 and Master Data List ref MDL002Z001 Issue A dated 29 April 2013. Operate in accordance with 328DO Flight Manual Supplement ref AFM016H043Z0001 Initial Release, dated 29 April 2013. Ma (...) | Aircraft Serial 1027. | 30.04.2013 |
| 10046552 | 328 DESIGN GMBH | | | | CL 604 VARIANT | Cabin refurbishment Incl LOPA change | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 328 Design Master Document List No. MDL006Z001 Issue A dated 25 September 2013. or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 30.09.2013 |
| 10014424 | A&M AVIONICS, SL | EASA.A.S.02469 | | | PIPER PA-31-350 | Verification of the B-RNAV Capability. Installation of GPS Modes Panel Annunciator. Installation of new XPDR Mode S Garmin GTX-330 and Encoder AK-350. | | A&M Avionics PC No. 02/2006 Rev. 0, dated 30/11/2005. | In accordance with Supplement No. SMV 02/06 R0, to the original Flight Manual approved by FAA. SN 31-7952088 | 21.12.2007 |
| 10014434 | A&M AVIONICS, SL | EASA.A.S.02483 | | | CESSNA 404 | Remove Weather Radar Bendix King RDS81. Install Weather Radar Bendix King DRD 2000 with MFD Avidyne EX 500. Reinvestigation of Dual GNS530 Installation according to A&M Avionics BIN° 017/2006 for B-RNAV Approval. Model: Cessna 404, S/N 404 0614 | | 2. Design Definition A&M Doc. DDI N° 03/06 Rev.1.2. New Supplements to Aircraft Flight Manual A&M Avionics Doc. SMV N° 03/06 Rev.1 and A&M Avionics Doc. SMV N° 04/06 Rev.1.2. Operating & Service Instructions: A&M Avionics Doc. BI N° 24/05 Rev.0 | 1. The design changes to be approved are only applicable to Cessna 404 airplanes with Dual GNS530 Installation according to A&M Avionics BI N° 017/2006, and EHSI SANDEL SN 3500 previously installed and approved as Primary Navigation Display. 2. P (...) | 30.01.2009 |
| 10014788 | A&M AVIONICS, SL | EASA.A.S.02988 | | | FAIRCHILD SWEARINGEN SA 226 TC | Installation of New Avionics Equipment. Transponders Mode S Filter Electronics TR7800H. | | New Supplement to Aircraft Flight Manual A&M Avionics Document SMV 01/07 R0. Operating & Service Instructions A&M Avionics Engineering Bulletin BI 02/07 R0. | | 08.10.2007 |
| 10014877 | A&M AVIONICS, SL | EASA.A.S.03104 | | | CESSNA 500 | Installation of New Avionics Equipment. Transponder Mode S GARMIN GTX-330D. | | New Supplement to Aircraft Flight Manual A&M Avionics Document SMV N° 08/07 R0. Operating & Service Instructions A&M Avionics Engineering Bulletin BI N° 08/07 R0. | | 05.02.2008 |
| 10043289 | A&M AVIONICS, SL | EASA.A.S.03339 | REV. 1 | | PA-34-200 | Installation of Dual COM/NAV/GPS GARMIN GNS 430 & GNS 530 | The Certification Basis for the original product remains applicable to this certificate/ approval. | Technical Documents to be Approved by EASA:- SERVICE BULLETIN BI N°: 19/07 R1- SMV N° 19/07 R1. Airplane Flight Manual Supplement for IFR operations with Dual Garmin GNS 530 & Garmin GNS 430, or later revisions of the above listed documents approved by EASA. | Approval limited to Airplane Model PA-34-200, S/N 34-7250309 | 19.02.2013 |
| 10046100 | A&M AVIONICS, SL | | | | T310R | Installation of Dual System Garmin GNS530W in accordance with Installation Bulletin BI 05/13 | Other: 14 CFR Part 23 according to Certification Programme PC & MCU05/13 The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Bulletin A<(&<)-M Avionics IB 05/13 Supplement to AFM A<(&<)-M Avionics SMV 05/13 or later revisions of the above listed documents approved by EASA | Approval limited to Airplane CESSNA T310R, S/N 310R-1408 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introd (...) | 14.08.2013 |
| 10032066 | A.I.M.S. INTERNATIONAL Corp. | | | FAA ST02675AT | BD-700-1A10 | Sound Reduction Kit | EASA IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA SoC dated 30 September 2010. FAA STC No. ST02675AT dated 22 April 2004. AIMS International Corp engineering top-level drawing 53G2521522 Rev I/R dated 15 March 2003 and installation instruction document No. AIMS-INST-1 Rev I/R dated 16 April 2003 or I (...) | None | 06.10.2010 |
| 10045364 | A.S. AVIONICS SERVICES LTDA. | | | | EMB-135BJ | Installation of IWG-A4 Potable Water Treatment Unit on EMB-135BJ EASA validation of ANAC STC 2012S09-08 | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List ref. EMB135BJ-38-0001MD, Rev. D, dated 11/09/2012. Certification Plan ref. EMB135BJ-38-0001CP, Rev. B, dated 07/03/2012, supplemented by EASA Declaration of Compliance ref. EMB135BJ-38-0001EASA-DC, Rev. Original, dated 20/02/2013. In (...) | Applicable to EMB-135BJ S/N 14500997, 14501029 and on Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce n (...) | 19.06.2013 |
| 10044063 | A/C SYSTEMS L.L.C. | | | FAA STC SA03065CH-D | G58, 58, 58A, D055, E55, E55A | Installation of an A/C Systems Air-conditioning System | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Standard Aero Master Drawing List 1021376 Rev (I), dated 20 April 2012. Operation in accordance with FAA Approved Airplane Flight Manual Supplement number FTA-010-6 Rev (R) dated 23 April 2012. Maintain in accordance wit (...) | The installer must determine whether this design change is compatible with previously approved modifications. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previous (...) | 14.03.2013 |



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| 10044069 | A/C SYSTEMS L.L.C. | | | FAA STC SA00257BO | 1- G36 2- F33A, A36 | Installation of an A/C Systems Air-conditioning System. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with A/C Systems Master Drawing List FTA-000-1 Rev C, dated 07 June 2012. Operation in accordance with FAA Approved Airplane Flight Manual Supplement number FTA-010-2 Rev (I) dated 25 April 2012. Maintain in accordance with A/ (...) | Model F33A Applicability Series CE-748, CE-772 and after Model A36 Applicability Series E1111, E-1241 through E-3635, excluding E-3630. Model G36 All Series. The installer must determine whether this design change is compatible with previously approved (...) | 15.03.2013 |
| 10015780 | AAE LTD | EASA.IM.A.S.01826 | REV. 1 | FAA STC ST01114WI | BOEING 737 CLASSIC SERIES | Installation of Smoke Detection and Fire Suppression System | | AAE Ltd. Master Drawing List TFA10-0061, Revision H, dated 01 December 2005 or later approved revision. AFM Supplement ref. TFA09-0063, TFA09-0064, TFA09-0065, TFA09-0066 dated 20 November 2002, TFA09-0085, TFA09-0086 dated 29 June 2005, TFA09-0088 (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 26.11.2002 |
| 10015840 | AAE LTD | EASA.IM.A.S.01934 | | FAA STC ST03001AT | FOKKER F.28 MARK 0100 | Smoke detection and Fire Suppression System for lower Deck Class C Cargo Compartments of Fokker F100 Series Aircraft | | 1. TFA 18-0179 Instructions for Continued Airworthiness. 2. TFA 10-0062 Master Drawing List. 3. TFA 18-0178 Structural Substantiation Report. 4. TFA 18-0134 Alteration Certification Plan. 5. TFA 18-0180 Weight Information Report. 6. TFA 18-0183 Alteration Com (...) | 1. AAE, Ltd. Document TFA09-0087, Revision IR, dated January 14, 2005 "FAA approved Airplane Flight Manual Supplement for the Class "C" Cargo Compartment Smoke Detection and Fire Suppression System on the Fokker F.28 Mark 0100 Series Aircraft" | 25.09.2006 |
| 10043324 | AAE LTD | EASA.IM.A.S.03172 | REV. 1 | FAA STC ST01077WI | A320 | Installation of AAE's smoke detection and fire suppression system in the lower Class C Cargo compartments. | | - FAA STC ST01077WI- Master Drawing List, ref. doc. TFA09-0057, Revision U, dated 11th March 2009 - Airplane Flight manual Supplement ref. doc. TFA09-0062/ doc. TFA09-0088/ doc. TFA09-0089, Revision IR, dated 4th November 2008 - Instruction for Contin (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The (...) | 23.06.2009 |
| 10040411 | AAR AIRCRAFT SERVICES | | | FAA STC ST03119CH | B717-200 | Interior Reconfiguration | EASA.IM.A.211 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged (...) | FAA STC No. ST03119CH dated 25 June 2012. AAR Engineering Services Master Data List ERP 11-062 Revision C dated 23 June 2012. Instructions for Continued Airworthiness (ICA), AAR Aircraft Services Report No. ERP 11-068 Revision IR, dated 28 March 2012 or (...) | None | 02.07.2012 |
| 10015931 | AAR AIRCRAFT SERVICES INC. - INDIANAP | EASA.IM.A.S.02068 | | SA7467SW | FOKKER F.27 | Installation of dual freon air conditioning systems in accordance with AAR Oklahoma, Inc., Drawing List 19L002, Revision F, dated July 19, 1988, or later FAA approved revision. | | AAR Oklahoma, Inc., Drawing List 19L002, Revision F, dated July 19, 1988, or later FAA approved revision. | 1. FAA approved flight manual supplement dated August 10, 1988, is required. 2. Compatibility of this modification with previously installed equipment must be determined by installer. | 13.10.2006 |
| 10015627 | AAR CARGO SYSTEMS | EASA.IM.A.S.01579 | REV. 1 | FAA STC ST02021CH | MD-11 | Installation of a Main Deck Cargo Handling System (FAA STC ST02021CH) | The Certification Basis for the original product remains applicable to this certificate/ approval | AAR Report 1826 Instruction for Continued Airworthiness | None | 10.12.2010 |
| 10028433 | AAR CARGO SYSTEMS | P- EASA.IM.A.S.03180 | | FAA STC ST02767CH | B767-300 | Install Main Deck Cargo Loading System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Certification basis is as defined in EASA STC 10028430. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Drawing List Report Number 3255, Revision C, 30/09/2009. Applicable revisions or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | To be installed on a/c converted by IAI Freighter Conversion (EASA STC10028430) | 08.01.2010 |
| 10040510 | AAR CARGO SYSTEMS | | REV. 1 | FAA STC ST02665CH | A300B4-601, A300B4-603 A300B4-605 R, A300B4-620 A300B4-622, A300B4-622 R | Validation of FAA Supplemental Type Certificate (STC) ST02665CH. Install a Main Deck Cargo Loading System (AAR Part Number 58535-101) on Airbus A300-600 series aeroplanes in accordance with AAR Master Drawing List Report Number 2149 Revision B, dated 15 December 2008. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AAR Weight & Balance Report A300-600 FSI Main Deck Cargo Loading System (MDCLS) P/N 58535-101 No. 2147 Rev. A, dated 25 Aug. 2008. AAR Instructions For Continued Airworthiness A300-600 FSI MDCLS P/N 58535-101 No. 2148 original issue, dated 16 July 2007 or (...) | none | 13.07.2012 |
| 10016873 | AAR COMPOSITES, INC. | EASA.IM.R.S.01249 | | FAA STC SR01934LA | SIKORSKY S-92A | Installation of Partial Search and Rescue Interior Configuration. | | Install in accordance with AAR Composites Master Data List SAR-620-010, Revision M dated 1/31/2007, or later approved revisions. Maintenance in accordance with AAR Composites Document Number AAR-ER-1169, Sikorsky S-92A SAR Search and Rescue (SAR) In (...) | No passengers permitted in the main cabin until the complete interior arrangement has been EASA or cognizant National Airworthiness Authority approved. | 07.05.2007 |
| 10016932 | AAR COMPOSITES, INC. | EASA.IM.R.S.01318 | | FAA STC SR03370AT | SIKORSKY S-92A | Structural provisions for a composite ramp floor panel. | | Install in accordance with AAR Composites Master Data List SRP-620-010, Revision B dated 18 December 2006, or later approved revisions. Maintenance in accordance with AAR Composites Document Number AAR-ER-1180, Revision A, dated December 7, 2006 or later approved revision. | MDL items under this STC meet Part 29.853. The installer is responsible for panel sealant and all fasteners as well as for structural provisions to the aircraft ramp door. User must not exceed the maximum load per surface area of the ramp panel as specified by the TC holder. | 07.05.2007 |
| 10017082 | AAR COMPOSITES, INC. | EASA.IM.R.S.01540 | | FAA STC NO. SR03077 AT | SIKORSKY S-92A | S-92A Structural Provisions for a Composite Passenger Cabin/VIP Interior. | | # S92-620-011 revision L dated 22 February 2008 or later approved revision # Master Document List# AAR-ER-1116 revision E dated 25 January 2008 or later approved revision # Instructions for Continued Airworthiness# AAR-ER-1264 revision IR da (...) | 1. The installer is responsible for the addition of decorative finish to continue to comply with JAR 29.853. AAR bulkheads are for interior partitions only and are not intended means of compliance for cargo or smoke retention that the installer must (...) | 23.04.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|--|---|--|--|--|---------------|
| 10015303 | AB LUFTFAHRTTECHNIK BERLIN GMBH | EASA.A.S.03868 | | | AIRBUS A319-112 | Change of the Cabin Configuration (150 Pax)/The cabin configuration will be changed to increase the certified 145pax configuration to a 150 pax configuration with minimum CabinCrew of 3 Flight Attendants | | # Certification Report, ref. doc. A319-CR25-03-2009, Issue 1, dated 17th March 2009# Design Modification Document, ref. doc. A319-DMD25-03-2009, Issue 1, dated 17th March 2009# Compliance Record Document, Issue 2, dated 5th May 2009 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 05.05.2009 |
| 10013958 | ACC COLUMBIA JET SERVICE GMBH | EASA.A.S.00516 | REV. 1 | | BOMBARDIER CL-600-2B16 (604) | Installation of an Artex ELT C406-1 | | Master Document List CL604-40163-000-S-LP | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 19.01.2005 |
| 10014138 | ACC COLUMBIA JET SERVICE GMBH | EASA.A.S.01249 | | | LEARJET 35 LEARJET 35A | Installation of Honeywell MST-67A dual Mode S XPDR/CAS-67A TCAS II System. | | - ACC Master Document List ACC-LR35-50096, revision 0, or later EASA-approved revisions. - ACC AFMS LR35-50096-820-S-MF, Revision 0, or later EASA-approved revisions. - ACC AFMS LR35-50096-830-S-MF, Revision 0, or later EASA-approved revisions. | | 28.04.2005 |
| 10014160 | ACC COLUMBIA JET SERVICE GMBH | EASA.A.S.01292 | | | FOKKER F-27 MARK 500 | Installation of an ARTEX ELT C406-1/2 | | | | 26.07.2005 |
| 10014764 | ACC COLUMBIA JET SERVICE GMBH | EASA.A.S.02954 | | | LEARJET 35 LEARJET 35A LEARJET 36 LEARJET 36A | Installation of Honeywell EGPWS Mk. VIII (in combination with an existing Weather Radar display) | | Master Document List LR35-60514-000-LP, rev. 1AFM Supplement LR35-60514-800-MF, rev. 1 | None | 02.03.2007 |
| 10017178 | ACC COLUMBIA JET SERVICE GMBH | EASA.R.S.01296 | | | EUROCOPTER DEUTSCHLAND EC-135- | Installation of a Helimap.MIDI.P0252 Moving Map System | | - Rotorcraft Flight Manual Supplement P1, P2, T1, (CDS, CPDS), EC 135-69008-800-S-MF Rev. 0 - Instructions for Continued Airworthiness EC 135-69008-810-S-MM Rev. 0 | | 19.09.2006 |
| 10017179 | ACC COLUMBIA JET SERVICE GMBH | EASA.R.S.01297 | | | EUROCOPTER DEUTSCHLAND MBB-BK | Installation of a Helimap MIDI.P0252 Moving Map System | | - Rotorcraft Flight Manual Supplement BK117 B-2, BK 117-69007-800-S-MF - Instructions for Continued Airworthiness BK117-69007-810-S-MM | | 19.09.2006 |
| 10017213 | ACC COLUMBIA JET SERVICE GMBH | EASA.R.S.01369 | | | EUROCOPTER DEUTSCHLAND MBB-BK | Installation of an EAE Heli.Map.MIDI.P0252 system with extension box (for advisory only). Installation of a moving map system into BK117B-2 and BK117C-1 models. | | Rotorcraft Flight Manual Supplement Eurocopter BK117B-2/C-1ACC-BK117-79003-800-S-MF Rev. 1, dated 31.05.2007. Instructions for Continued Airworthiness ACC-BK117-79003-810-S-MM Rev.0, dated 03.04.2007. | | 14.06.2007 |
| 10034112 | ACC COLUMBIA JET SERVICE GMBH | | | | CL-600-2B16, CL-600-2B19 | Installation of a combined IRIDIUM-GPS Antenna as provision of optional Sky Tracking System. | EASA IM-A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | ACC Master Document List ACC-CRJ200-09016 Revision 0 dated 24 February 2011. ACC ICA CRJ200-09016-810-MM Revision 0 dated 29 November 2010, or later revisions of the above listed documents approved by EASA. | None | 08.03.2011 |
| 10014653 | ACI SYSTEMES | EASA.A.S.02810 | | | BEECH 200 | ATC replacement and Mode S installation. | | ACI S007/2006AFMS No ACI 05 AFM S007 Revision 0 or later approved revision | AFMS must be on board when modified | 07.05.2007 |
| 10014737 | ACI SYSTEMES | EASA.A.S.02920 | | | CESSNA 550 | Single pilot operation and associated modifications | | AFM Supplement ACI-00-AFM-M-021-Rev. 1 Compliance Checklist ACI-00-CCL-M-021-Rev. 3, dated 05/06/2007 | | 06.06.2007 |
| 10014902 | ACI SYSTEMES | EASA.A.S.03137 | | | REIMS AVIATION CESSNA F 406 | ACI-024 - Universal UNS-1L Installation - B-RNAV and P-RNAV Capability. | | Flight Manual Supplement ACI-10-AFM-M-024-Rev3, Dated 14 September 2007. | F406 equipped with KCP220 Allied Signal AutoPilot. | 10.10.2007 |
| 10014905 | ACI SYSTEMES | EASA.A.S.03140 | | | REIMS AVIATION CESSNA F 406 | ACI-030 - Installation of an NAT NTX138 - FM Communication device. | | - | | 08.10.2007 |
| 10014996 | ACI SYSTEMES | EASA.A.S.03293 | | | BEECH 90 | GNSS 430 and B-RNAV capability on Beech 90 | | AFM Supplement ACI-09-AFM-M-047-Rev0 | N/A | 18.02.2008 |
| 10015003 | ACI SYSTEMES | EASA.A.S.03303 | | | FALCON 50 FALCON 900 | Falcon 50 and F900 EHS Mode S Capability | | AFM Supplement ACI-05-AFM-M-033-Rev0 ACI-05-TD-M-033-Rev0 | Applicability for Falcon 50 and F900 which are already Mode A, C and ELS Mode S certified, ACAS certified and equipped with a dual transponder system. | 29.10.2007 |
| 10015016 | ACI SYSTEMES | EASA.A.S.03320 | | | CESSNA 550 | ACI-045 - DFDR Upgrade. | | AFM Supplement ACI-12-AFM-M-045-Rev. 1 Maintenance Manual Supplement ACI-12-MPS-M-045-Rev.0 | This STC is only applicable to aircraft equipped with Fairchild F800 FDR. | 16.05.2008 |
| 10015075 | ACI SYSTEMES | EASA.A.S.03406 | | | BEECH 200 BEECH 300 | Installation of a Garmin 400 Series with B-RNAV capability on Beech 200. | | | | 05.02.2008 |
| 10015099 | ACI SYSTEMES | EASA.A.S.03441 | | | BEECH 90 | Installation of Garmin 500 Series GPS with B-RNAV capability on Beech 90. | | ACI-09-AFM-M-070, Revision 0, dated 15 January 2008 Modification file ACI-09-TD-M-070, Revision 0, dated 15 January 2008. | The supplemental Airplane Flight Manual ACI-09-AFM-M-070, revision 0, dated 15 January 2008 must be on board when the modification is installed. | 04.04.2008 |
| 10016968 | ACROHELIPRO GLOBAL SERVICES, INC | EASA.IM.R.S.01369 | | FAA STC SR02292AK | BELL 206 BELL 206A BELL 206B BELL 206L BELL 206L-1 BELL 206L-3 BELL 206L-4 | Sagem ICDS and Crossbow AHRs Installation. Consists of two 8 inch integrated cockpit display units (one PFD and one MFD) including engine instruments and an attitude and heading reference system. This STC is classified as "BASIC" by the FAA and is a (...) | | HP 206-34-33-102 # Certification Data List # revision A dated 3 July 2007 or later approved revision; HP 206-34-33-251 - Instructions for Continued Airworthiness - revision N/C dated 3 February 2007 or later approved revision; Addendum to ICA HP 206-34 (...) | Maximum ambient operating temperature of 40 degrees Celsius (103 degrees Fahrenheit) | 31.10.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------|---------------------|----------------|--------------------------------|------------------------|--|--|---|---|---------------|
| 10016665 | ACS-NAI Ltd. | EASA.IM.A.S.03135 | REV. 1 | TCCA LSTC C-LSA09-031/D ISS. 2 | BOMBARDIER CL-600-2B19 | Installation, Complete Custom Aircraft Interior | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA LSTC No. C-LSA09-031/D Issue 2 dated 17 June 2013. Aero Consulting Services Ltd. Modification Data Summary G120000 Revision H. Midcoast Aviation Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Installation, 33-299M00 (...) | Restricted to ASN 8088. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 16.08.2013 |
| 10016669 | ACS-NAI Ltd. | EASA.IM.A.S.03151 | REV. 1 | LSTC C-LSA09-009/D ISSUE 2 | CL-600-2B19 | Installation, Complete Custom Aircraft Interior | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada LSTC C-LSA09-009/D Issue 2 dated 28 September 2012. Aero Consulting Services Ltd. Modification Data Summary G099000 Rev E. Midcoast Aviation Inc. Flight Manual Supplement 33-299M006 Rev NC or later revisions of the above listed docume (...) | Restricted to ASN 8081. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 24.05.2013 |
| 10026769 | ACS-NAI Ltd. | P-EASA.IM.A.S.03137 | REV. 1 | TCCA LSTC C-LSA09-026, ISS. 1 | CL-600-2B19 | Installation, Complete Custom Aircraft Interior | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA LSTC No. C-LSA09-026 Issue 1 dated 04 August 2009. Aero Consulting Services Ltd. Modification Data Summary G103000 Revision G. Midcoast Aviation Inc./ Aero Consulting Services Ltd Airplane Flight Manual Supplement for the Complete Custom Aircraft In (...) | Restricted to ASN 8079. | 16.08.2013 |
| 10026840 | ACS-NAI Ltd. | | REV. 1 | TCCA LSTC C-LSA09-028/D, ISS.1 | BOMBARDIER CL-600-2B19 | Installation, Complete Custom Aircraft Interior. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA LSTC No. C-LSA09-028/D Issue 1 dated 04 August 2009. Aero Consulting Services Ltd. Modification Data Summary G100000 Revision E. Midcoast Aviation Inc./ Aero Consulting Services Ltd Airplane Flight Manual Supplement for the Complete Custom Aircraft I (...) | Restricted to ASN 8070. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 16.08.2013 |
| 10026904 | ACS-NAI Ltd. | P-EASA.IM.A.S.03136 | REV. 1 | TCCA LSTC C-LSA09-032/D | BOMBARDIER CL-600-2B19 | Installation, Complete Custom Aircraft Interior. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA LSTC No. C-LSA09-032/D Issue 1 dated 12 August 2009. Aero Consulting Services Ltd. Modification Data Summary G121000 Revision D. Midcoast Aviation Inc Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Installation, 33-299M (...) | Restricted to ASN 8089. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 16.08.2013 |
| 10027086 | ACS-NAI Ltd. | P-EASA.IM.A.S.03165 | REV. 2 | TCCA LSTC NO. C-LSA09-033/D | CL-600-2B19 | Installation, Complete Custom Aircraft Interior | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA LSTC No. C-LSA09-033/D Issue 2 dated 24 September 2010. Aero Consulting Services Ltd. Modification Data Summary G122000 Revision F. Midcoast Aviation Inc Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Installation, 33-2 (...) | Restricted to ASN 8085. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 16.08.2013 |
| 10028761 | ACS-NAI Ltd. | P-EASA.IM.A.S.03154 | REV. 1 | TCCA LSTC C-LSA09-004D, ISS. 2 | BOMBARDIER CL-600-2B19 | Installation, Complete Custom Aircraft Interior | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA LSTC No. C-LSA09-004D Issue 2 dated 22 July 2009. Aero Consulting Services Ltd. Modification Data Summary G115000 Revision B. Savannah Air Center Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior, G115090 Revision A, (...) | Restricted to ASN 8083. | 16.08.2013 |
| 10028762 | ACS-NAI Ltd. | P-EASA.IM.A.S.03131 | REV. 1 | TCCA LSTC NO. C-LSA09-023/D | BOMBARDIER CL-600-2B19 | Installation, Complete Custom Aircraft Interior. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA LSTC No. C-LSA09-023/D Issue 1 dated 02 July 2009. Aero Consulting Services Ltd. Modification Data Summary G098000 Revision D. Midcoast Aviation inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Installation, 33-299M00 (...) | Restricted to ASN 8065. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 16.08.2013 |
| 10028763 | ACS-NAI Ltd. | P-EASA.IM.A.S.03134 | REV. 1 | TCCA LSTC NO. C-LSA09-030/D | BOMBARDIER CL-600-2B19 | Installation, Complete Custom Aircraft Interior. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA LSTC No. C-LSA09-030/D Issue 2 dated 21 July 2009. Aero Consulting Services Ltd. Modification Data Summary G119000 Revision C. Midcoast Aviation Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Installation, 33-299M01 (...) | Restricted to ASN 8078. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 16.08.2013 |
| 10031149 | ACS-NAI Ltd. | | | TCCA STC SA10-26 | CL-600-2B19 | Installation, CMC Electronics Pilot view Class II Electronics Flight Bag | EASA.IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA SoC ref. C-10-0364 dated 20 July 2010. TCCA STC ref. SA10-26 Issue 1 dated 19 March 2010. ACS-NAI Ltd Modification Data Summary Number G370000 Revision A. Savannah Air Center Airplane Flight Manual Supplement for Class 2 Electronic Flight Bag (EFB), (...) | None | 29.07.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------|-------------------|----------------|--------------------------------|---|---|--|---|---|---------------|
| 10031944 | ACS-NAI Ltd. | | | TCCA STC #C-LSA10-034/D ISS 1 | BD-700-1A10 | Installation of Two 16G Side Facing Divans | EASA TCDS No EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada SoC ref. C-10-0708 dated 22 September 2010. Transport Canada STC No. C-LSA10-034/D Issue 1 dated 16 August 2010. ACS-NAI Ltd. Modification Data Summary Number G471000 Revision NC or later approved revision. Refer to drawings listed in (...) | Restricted to ASN 9347 | 23.09.2010 |
| 10032295 | ACS-NAI Ltd. | EASA.IM.A.S.03152 | REV. 2 | TCCA STC C-LSA09-021/D ISS.3 | CL-600-2B19 | Installation, Complete Custom Aircraft Interior | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. C-LSA09-021/D Issue 3 dated 17 June 2013. ACS-NAI Ltd. Modification Data Summary G116000 Revision F. Aero Consulting Services Ltd. Mandatory Maintenance Manual Supplement G116070 Revision NC. Savannah Air Center Inc. Airplane Flight Manual (...) | Restricted to ASN 8080 | 16.08.2013 |
| 10032519 | ACS-NAI Ltd. | | | TCCA STC NO. SA10-25 | DHC-8-401, DHC-8-402 | Installation of LED Lavatory Light System | JAR/CS-25 as per EASA.IM.A.191The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. plus: Special Condition H-01 "Enhanced Airworthiness Prog (...) | Installation must be in accordance with ACS-NAI Ltd. TCCA approved Modification Data Summary Number E 998000, Rev. N/C. Maintenance must be in accordance with Instructions for Continued Airworthiness 998071, Rev. A, E 998073 Rev. A and E 998074, (...) | none | 12.11.2010 |
| 10032736 | ACS-NAI Ltd. | EASA.IM.A.S.03130 | REV. 2 | TCCA LSTC C-LSA09-024/D, ISS.3 | BOMBARDIER CL-600-2B19 | Installation, Complete Custom Aircraft Interior | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA LSTC No. C-LSA09-024/D Issue 3 dated 20 October 2010. Aero Consulting Services Ltd. Modification Data Summary G102000 Revision G. Midcoast Aviation Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Installation, 33-29 (...) | Restricted to ASN 8077. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 16.08.2013 |
| 10033800 | ACS-NAI Ltd. | | | TCCA STC NO.C-LSA11-040/D | BD700-1A11 | Cabin Modifications for Wheelchair Access | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged (...) | TCCA SoC ref. C-10-1061 dated 08 February 2011 TCCA STC No. C-LSA11-040/D Issue 1 dated 02 February 2011 ACS-NAI Ltd. Modification Data Summary G950000 Revision NC or later approved revision or later revisions of the above listed documents approved by E (...) | Restricted to ASN 9373. Previous embodiment of EASA STC No. 10031088 (TCCA STC SA05-60). | 09.02.2011 |
| 10034703 | ACS-NAI Ltd. | | REV. 1 | TCCA STC SA07-40 | PC-12/45, PC-12/47, PC-12/47E | Installation Utility Door Existing Cargo door modified to incorporate a utility door (900 mm high by 635 mm wide) | FAR 23The Certification Basis for the original product remains applicable to this certificate/ approval. | Flight manual supplement Doc. No. F195090. Modification data summary Doc. No. F195000. Maintenance manual supplement Doc. No. E550070-MMS; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision). | none | 28.11.2012 |
| 10034865 | ACS-NAI Ltd. | | | TCCA STC SA98-60 ISSUE 2 | DHC-8-102,-103,-106 DHC-8-201,-202 DHC-8-301,-311,-314,-315 DHC-8-401,-402 | Flight Deck Manual Stowage Box TCCA STC SA98-60, Issue 2, dated 12.02.2001. | FAR 25 as per EASA.IM.A.191, Iss. 4 for DHC-8-100/200/300 Series. JAR-CS-25 as per EASA.IM.A.191, ISS, 4 for DHC-8-400 Series. The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation must be in accordance with ACS-NAI Ltd. TCCA approved Modification Data Summary B 78600, Rev. H (>,<,-) - Maintenance must be in accordance with Instructions for Continued Airworthiness MMS B 78671 (>,<,-) or later revisions of the above listed (...) | None. | 06.05.2011 |
| 10035310 | ACS-NAI Ltd. | | REV. 1 | TCCA STC C-LSA11-075/D ISS1 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. C-LSA11-075/D Issue 1 dated 13 May 2011. ACS-NAI Ltd. Modification Data Summary G789000 Revision NC or later approved revision. ACS-NAI Ltd. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior, G789091 Revision B or (...) | Restricted to ASN 9403. This system is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require re-examination of the cert (...) | 15.06.2011 |
| 10036101 | ACS-NAI Ltd. | | | TCCA STC C-LSA11-134/D | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9412 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged (...) | TCCA STC No. C-LSA11-134/D Issue 1 dated 03 August 2011. ACS-NAI Ltd. Modification Data Summary H039000 Revision NC or later approved revision. ACS-NAI Ltd. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior, G789091 Revision B (...) | Restricted to ASN 9412. The Wireless Local Area Network (WLAN) system is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require re-examination of the cert (...) | 11.08.2011 |
| 10036345 | ACS-NAI Ltd. | | | TCCA STC C-LSA11-058/D | BD-700-1A10 | Installation of Tailwind 500 System | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged (...) | -Transport Canada STC No. C-LSA11-058/ D Issue 1 dated 21 April 2011. ACS-NAI Ltd. Modification Data Summary H058000 Rev N/C or later approved revision. ACS-NAI Ltd Instructions for Continued Airworthiness H058071 Rev N/C or later approved revision. - Tran (...) | None | 02.09.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------|----------------------------|----------------|--------------------------------|--|--|--|--|--|---------------|
| 10037710 | ACS-NAI Ltd. | | | TCCA STC C-LSA10-169/D | CL-600-2B19 | Installation, Complete Custom Aircraft Interior. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. C-LSA10-169/D Issue 1 dated 03 December 2010. ACS-NAI Ltd. Modification Data Summary G773000 Revision NC or later approved revision. Aero Consulting Services Ltd. Mandatory Maintenance Manual Supplement G192070 Revision NC or later approved (...) | Restricted to ASN 8101. | 16.12.2011 |
| 10037711 | ACS-NAI Ltd. | | | TCCA STC C-LSA11-165/D | CL-600-2B19 | Installation, Complete Custom Aircraft Interior. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. C-LSA11-165/D Issue 1 dated 11 October 2011. ACS-NAI Ltd. Modification Data Summary H133000 Revision NC or later approved revision. Aero Consulting Services Ltd. Mandatory Maintenance Manual Supplement G192070 Revision NC or later approved r (...) | Restricted to ASN 8107. | 16.12.2011 |
| 10038022 | ACS-NAI Ltd. | | | TCCA STC NO. C-LSA11-009/D | CL-600-2B19 | Installation, Complete Custom Aircraft Interior. | EASA.IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. C-LSA11-009/D Issue 2 dated 22 July 2011. ACS-NAI Ltd. Modification Data Summary G583000 Revision A or later approved revision. ACS-NAI Ltd. Mandatory Maintenance Manual Supplement, C583070 Revision NC or later approved revision. ACS-NAI Ltd. (...) | Restricted to ASN 81051. The Wireless Local Area Network (WLAN) system is intended as part of their interior completion is intended to provide cached internet content and email services to Portable Electronic Devices. Any other use of this equipment will re (...) | 12.01.2012 |
| 10038149 | ACS-NAI Ltd. | | | TCCA STC NO. C-LSA11-179/D | BD-700-1A10 | Modifications to Existing Interior Completions ASN 9393. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. C-LSA11-179/D Issue 3, dated 12 January 2012. Aero Consulting Services Modification Data Summary H267000, Revision B, dated 22 December 2011 or later approved revision; Aero Consulting Services Mandatory Maintenance Manual Supplement for An (...) | Restricted to ASN 9393 and The Installation of EASA STC No. 1003660 is a prerequisite for the approval of this STC. | 24.01.2012 |
| 10039631 | ACS-NAI Ltd. | | | TCCA STC C-LSA05-318/D | CL-600-2B16 | Aircraft Completion ASN 5618, 12 Passenger Interior Configuration. | EASA.IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. C-LSA05-318/D Issue 2, dated 21 July 2006. Aero Consulting Services Ltd Modification Data Summary 610000 Revision A, dated 18 July 2006. Midcoast Aviation Airplane Flight Manual Supplement 34-424M005 Revision A. Midcoast Aviation Airplane FI (...) | Restricted to ASN 5618 | 15.05.2012 |
| 10039703 | ACS-NAI Ltd. | TCCA STC NO. C-LSA12-038/D | | | BD-700-1A10 | Installation of Complete Custom Aircraft Interior - ASN 9434 | EASA.IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. C-LSA12-038/D Issue 1, dated 30 April 2012. Aero Consulting Services Ltd Configuration Definition List H282000 Revision NC. Aeroplane Flight Manual Supplement H282090, Revision N/C. Instructions for Continued Airworthiness for the installation (...) | Restricted to ASN 9434. The Wireless Local Area Network (WLAN) system is intended to provide internet connectivity and email services to the airplane's cabin using portable devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis. | 18.05.2012 |
| 10040686 | ACS-NAI Ltd. | JAA STC ACG.21NE2.084 | | TCCA SA01-77/JAA ACG.21NE2.084 | DHC-8-102,-103,-106 DHC-8-201,-202, DHC-8-301,-311,-314,-315 | Installation of an improved Aft Jet Pipe Support TCCA STC N° SA01-77, Issue 3 | FAR 25 as per EASA.IM.A.191, Issue 5 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Must be in accordance with Aero Consulting Service Ltd. Modification Data Summary N° C 54600, Revision B. Maintenance: Must be in accordance with Aero Consulting Service Ltd. Maintenance Manual Supplement N° C 54672, N/C or later revisions o (...) | None. | 19.07.2012 |
| 10041474 | ACS-NAI Ltd. | | REV. 1 | TCCA STC SA12-17 | PC-6/B2-H2 PC-6/B2-H4 | Installation of Vapour Cycle Air Conditioner System. | US CAR 3 dated May 15, 1956 Amrdt: 3-1 to 3-5and Special Conditions notified by the FAA to the Swiss FOCA The Certification Basis for the original product remains applicable to this certificate/ approval. Derogation to latest amendment (Part 21 A.101); T (...) | Airplane Flight Manual Supplement H404091 Revision N/C; Maintenance Manual Supplement H404071 Revision N/C and H404072 Revision N/C; Wiring Diagram Manual Supplement H404073 Revision N/C; or later revisions of the above listed documents approved by EASA (...) | Refer to the Airworthiness Limitations contained in Airplane Flight Manual Supplement H404091 Revision N/C. | 04.10.2012 |
| 10042346 | ACS-NAI Ltd. | ACG.21NE2.108 | | TCCA SA02-127 / ACG.21NE2.108 | DHC-8-102,-103,-106 DHC-8-201,-202, DHC-8-301,-311,-314,-315 | Installation of an Improved Ground Power Current Limiter (ACS-NAI Ltd. Modification Data Summary D 11900) TCCA STC SA02-127, Issue 1 | FAR 25 as per EASA.IM.A.191, Issue 6 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation: Must be in accordance with ACS-NAI Ltd. Modification Data Summary D 11900, Revision C, dated 22.03.2012 and ACS-NAI Ltd. Installation Instructions D 11901, Rev. E, dated 23.03.2012. Maintenance: Must be in accordance with ACS Ltd. Main (...) | None. | 26.11.2012 |
| 10043074 | ACS-NAI Ltd. | EASA.IM.A.S.03133 | REV. 1 | TCCA STC C-LSA09-029/D, ISS.2 | CL-600-2B19 | Installation, Complete Custom Aircraft Interior | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA LSTC No. C-LSA09-029/D Issue 2 dated 17 June 2013. Aero Consulting Services Ltd. Modification Data Summary G118000 Revision D. Midcoast Aviation Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Installation, 33-299M00 (...) | Restricted to ASN 8087. | 16.08.2013 |
| 10043075 | ACS-NAI Ltd. | EASA.IM.A.S.03132 | REV. 1 | TCCA STC C-LSA09-027/D, ISS. 2 | CL-600-2B19 | Installation, Complete Custom Aircraft Interior | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA LSTC No. C-LSA09-027/D Issue 2 dated 17 June 2013. Aero Consulting Services Ltd. Modification Data Summary G117000 Revision H. Midcoast Aviation Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Installation, 33-299M00 (...) | Restricted to ASN 8084. | 16.08.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|--|--|--|---|---|---------------|
| 10043177 | ACS-NAI Ltd. | | | | CL-600-2B16 (604 VARIANT) | Installation, Aircell Communication System | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | -Transport Canada STC No. C-LSA12-046/D, Issue 1 dated 12 July 2012 -ACS-NAI Ltd Modification Data Summary H048000 Revision NC.-ACS-NAI Ltd Mandatory Maintenance Manual Supplement for the installation, Iridium Antenna FS 580, H048075 Revision N/C.-ACS (...) | Restricted to ASN 5541 | 08.01.2013 |
| 10044051 | ACS-NAI Ltd. | | | TCCA STC SA12-40 | BD-700-1A10, BD-700-1A11 | Installation, Complete Custom Aircraft Interior | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25 Amendment 1 Special Condition(s): CRI D-02 Issue 2, Side Facing Divans Equivalent Safety Finding(s): CRI D-04 Issue 2 (...) | Transport Canada STC No. SA12-40 Issue 1, dated 08 June 2012 ACS-NAI Configuration Definition Record H525001 Revision NC, dated 07 June 2012 ACS-NAI Airplane Flight Manual Supplement H525000 Revision NC Instructions for Continued Airworthiness document IC (...) | EASA approved configurations are identified in Configuration Definition Record H525001. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplanes cabin using portable electronic devices (PEDs (...) | 14.03.2013 |
| 10046554 | ACS-NAI Ltd. | | | TCCA C-LSA13-057/D | CL-600-2B19 | Cabin Interior Re-configuration | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. C-LSA13-057/D Issue 1 dated 8th September 2013. ACS-NAI Ltd. Modification Data Summary H989000 Revision N/C or later approved revision. ACS-NAI Ltd. Airplane Flight Manual Supplement H989001 Revision N/C or later approved revision (...) | Restricted to ASN 7248 | 30.09.2013 |
| 10015375 | ACSS, AVIATION COMMUNICATION AND | EASA.IM.A.S.01045 | | FAA STC ST01583LA | BOEING 737-300 BOEING 737-400 | Installation of ACSS T2CAS System according to FAA STC ST01583LA dated 20th January 2004.- ACSS Index of Master Data Lists, Document No. 8900034, Revision "A" dated 3rd September 2004, or later FAA approved revision- ACSS Airplane Flight Manual Supp (...) | - ACSS Index of Master Data Lists, Document No. 8900034, Revision "A" dated 3rd September 2004, or later FAA approved revision- ACSS Airplane Flight Manual Supplement, Document No. 8002391-001, Revision "A" dated 21st October 2004, or later FAA approved revision | 1. According to limitations section in FAA STC ST01583LA dated 20th January 2004.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other a (...) | 17.08.2005 | |
| 10015406 | ACSS, AVIATION COMMUNICATION AND | EASA.IM.A.S.01117 | | FAA STC ST01702LA | BOEING 747-200 BOEING 747-300 | Installation of ACSS T2CAS | -in accordance with Approval Report DL/LVT/05.532542-FAA approved ACSS Index of Master Data lists, Document No 8900035 revision A, dated 27 October 2004 or later approved revision-FAA approved ACSS Airplane Flight Manual Supplement Document No 8003638- | 1. in accordance with Approval Report DL/LVT/05.5325422. T2CAS Airplane Flight Manual Supplement (AFMS), ACSS document number 8003638-001 revision A, approved by the FAA on May 25, 20053. This STC is approved only for the product configuration as de (...) | 01.08.2005 | |
| 10015573 | ACSS, AVIATION COMMUNICATION AND | EASA.IM.A.S.01480 | | FAA STC SA4839NM | BOEING 757-200 | Installation and activation of TCAS and the XS-950 Mode S Transponder (including ELS/EHS). | Honeywell Master Data List ML4070373 Revision F dated July 24 1990 or later approved revision. Honeywell AFM Supplement 5140-0885 approved July 14 1990 or later approved revision. | 07.11.2005 | | |
| 10015650 | ACSS, AVIATION COMMUNICATION AND | EASA.IM.A.S.01625 | | FAA STC SA5061NM | DC-9-81 (MD-81) DC-9-82 (MD-82) DC-9-83 (MD-83) DC-9-87 (MD-87) MD-88 | Installation of Honeywell Traffic Alert & Collision Avoidance System (TCAS II) | -Honeywell Master Data List No ML40741700 Revision A dated March 21 1991 or later approved revision-Honeywell AFM supplement TCAS-APPS-02494 Revision A dated October 18, 2000 or later approved revision | 30.01.2006 | | |
| 10015651 | ACSS, AVIATION COMMUNICATION AND | EASA.IM.A.S.01626 | | FAA STC SA5541NM | BOEING 767-200 BOEING 767-300 | Installation of Honeywell Traffic Alert & Collision Avoidance System (TCAS) | Honeywell Master Data List No ML4076520 Revision N/C dated July 24 1992 or later approved revision | 1. TCAS Airplane Flight Manual Supplement (Change 7) TCAS-APPS-0294, Rev A.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft (...) | 12.12.2005 | |
| 10015731 | ACSS, AVIATION COMMUNICATION AND | EASA.IM.A.S.01753 | | - | AIRBUS A320-200 | Installation of Honeywell TCAS II system | Master Data List ref. ML-4075160, revision A, dated September 25, 1992 or later approved revision AFM supplement ref. TCAS-APPS-0294 revision A or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.11.2005 | |
| 10015838 | ACSS, AVIATION COMMUNICATION AND | EASA.IM.A.S.01932 | | FAA STC SA4882NM | BOEING 747-100 BOEING 747-100B SUD BOEING 747-200B BOEING 747-200C BOEING 747-200F BOEING 747-300 BOEING 747-400 BOEING 747-SP BOEING 747-SR | Installation of a Honeywell Traffic Alert and Collision Avoidance System (TCAS II) in accordance with FAA approved Honeywell Master Data List number ML4071598 | Data package containing the following documents, and additional supporting data: 1. FAA Letter to Honeywell of approval pertaining to Honeywell previously approved STCs TCAS II installations to be upgraded to TCAS (10004) and to Mode S (10004) as (...) | FAA approved Flight Manual Supplement required for this configuration is listed on the appropriate approved Honeywell and ACSS Master Index List. | 31.07.2006 | |
| 10031364 | ACSS, AVIATION COMMUNICATION AND | | | FAA STC ST01911LA | B747-400 | Installation of ACSS Mode S Transponder with ELS/EHS capability to replace Mode S Transponder in existing ACSS Traffic Alert and Collision Avoidance System TCAS II in accordance with FAA approval ACSS Master Data List Document No. 8900019-001, Revision (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. | None | See Limitation Section of FAA approved ACSS Master Data List No. 8900019 | 11.08.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|-----------------------|---|--|--|--|--|---------------|
| 10032094 | ACSS, AVIATION COMMUNICATION AND | | | FAA STC NO. ST00411LA | B747-400 | Installation of Honeywell Traffic Alert and Collision Avoidance System(TCASII) | The Certification Basis for the original product remains applicable to this certificate/ approval. | FAA approved Air New Zealand Master Drawing List TX30031 Revision A, dated June 26, 1996 or later approved revision. For appropriate FAA approved Airplane Flight Manual Supplement refer to associated section of the Air New Zealand Master Drawing List Document. | none | 07.10.2010 |
| 10042297 | ACSS, AVIATION COMMUNICATION AND | EASA.IM.A.S.01059 | REV. 1 | FAA ST01646LA | ATR 42-200, ATR 42-300 ATR 42-320, ATR 42-500 ATR 72-101, ATR 72-102 ATR 72-202, ATR 72-211 ATR 72-212, ATR 72-212A | Installation of ACSS Terrain and Traffic Collision Avoidance System(T2CAS).The revision 1 has introduced an AFM Supplement update. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ACSS Master Data List document, ref: 8900038 Rev. BEASA Aircraft Flight Manual Supplement, ref: 8004809-001 Rev. Aor later revisions of the above listed documents approved by EASA | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 23.01.2013 |
| 10042553 | ACSS, AVIATION COMMUNICATION AND | EASA.IM.A.S.01044 | REV. 1 | FAA STC SA5530NM | B737-100/200/300/400/500 | FAA STC SA5530NM validation - Installation of the ACSS TCASII System with Change 7.1 software | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ACSS Index of Master Data Lists (IMDL), Reference 8900010, Revision J, FAA approved the 06 July 2010;- Airplane Flight Manual Supplement (AFMS) according to Honeywell Master Data List (MDL), Reference ML4075161, Revision AH, dated 14 May 2002;- ACSS (...) | None. | 14.12.2012 |
| 10042711 | ACSS, AVIATION COMMUNICATION AND | | REV. 3 | FAA STC ST02406LA | SEE EASA APPROVED MODEL LIST | Installation of Traffic Alert and Collision Avoidance Systems (TCAS II) version 7.1. Revision to incorporate additional aircraft models to the Approved Model List: 1. Hawker Beechcraft Corporation: BAe 125 Series 1000A, BAe 125 Series 800B, and Hawker 800X (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC ST02406LA, issued 6th February 2012-FAA Approved Model List AML No. ST02406LA, dated 3rd June 2013-Master Data List ref. doc. 8900067-002, Rev. D, dated 16th May 2013-Aircraft Flight Manual Supplement ref. doc. 8008565-001, Rev. IR, dated 9th (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 15.10.2013 |
| 10042940 | ACSS, AVIATION COMMUNICATION AND | EASA.IM.A.S.01064 | | FAA STC ST02788AT | B767-200/-300/-300F | Installation of Provisions for Terrain and Traffic Collision Avoidance System (TTCAS) This STC is a validation of FAA STC ST02788AT. | | EMTEQ Engineering Master Drawing List EMTEQ-025-MDL Revision dated January 27, 2004 or later FAA approved revision-AFM supplement Doc No 8002964-001 | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 09.03.2005 |
| 10042943 | ACSS, AVIATION COMMUNICATION AND | EASA.IM.A.S.01065 | REV. 1 | FAA STC ST001627LA | B767-200/-300 | Installation of T2CAS | | -FAA STC ST001627LA-Doc. No. 8900037, Rev A dated 11 May 2004 or later FAA revision -AFM Doc. No. 8002964-001, Rev "A" dated 27 April 2004, or later FAA revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 21.09.2005 |
| 10042944 | ACSS, AVIATION COMMUNICATION AND | EASA.IM.A.S.01067 | REV. 1 | FAA STC ST001628LA | B757-200 | Installation of T2CAS | | -FAA STC ST001628LA-ACSS Index of Master Data Lists Document Number 8900036 Revision B dated May 10 2004 or later FAA approved revision-ACSS Flight Manual Supplement number 8002942-001 revision "A" dated May 5 2004 or later FAA approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 21.09.2005 |
| 10044749 | ACSS, AVIATION COMMUNICATION AND | | | FAA STC SA4839NM | 757-200, 757-200PF | Installation of the ACSS TCAS system (description: Installation and activation of ACSS TCAS Change 7.1 software and deactivation of the ACSS TCAS Change 7.0 software) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Data List, Honeywell document No. ML4070373, Revision F, dated 24 July 1990 - Aircraft Flight Manual Supplement, Honeywell document No. 5140-0885, approved on 14 July 1990; or later revisions of the above listed documents approved by EASA in (...) | None | 03.05.2013 |
| 10033485 | ADAC LUFTFAHRT TECHNIK GmbH | | | | EC135P2 (CPDS), EC135 P2+, EC135T2 (CPDS), EC135 T2+ | Installation of Dual Garmin GNS 430W into EC 135 with MEGHAS cockpit installation of two GNS430W linked with indication on the FCDSS as defined in the Master Drawing List EC135-01 066-005-LIN and based on ALT Major Change EC135-01066. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Flight Manual Supplement:-, EC135-01066-025-MFS-00 Rev. 0 or later revisions of the above listed documents approved by EASA, -EC135-01066-023-MMS Rev. 0 Instructions for Continued Airworthiness, -EC135-01066-004-DIN Rev. 0 Installation Instructions or later revisions of the above listed documents. | None | 25.01.2011 |
| 10035671 | ADAC LUFTFAHRT TECHNIK GmbH | | | | EC135 T1(CDS) EC135 T1(CPDS) EC135 T2(CPDS) EC135 T2+ EC135 P1(CDS) EC135 P1(CPDS) EC135 P2(CPDS) EC135 P2+ EC135 P2+ | Installation of fixed Landing Lights on the forward cross tube. Installation of a second fixed landing light on the forward cross tubes defined in the MDL EC135-01072-001-LMD and based on ALT change project EC135-01072. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement:-, EC135-01072-025-MFS-00 Rev. 0 or later revisions of the above listed documents approved by EASA, -EC135-01072-023-MMS Rev. 0 Instructions for Continued Airworthiness (-, <-), <-> EC135-01072-004-DIN Rev. 0 Installation Instructi (...) | None | 08.07.2011 |
| 10036349 | ADAC LUFTFAHRT TECHNIK GmbH | | REV. 1 | | EC135 P2+ / EC135 P2(CPDS) EC135 T2+ / EC135 T2(CPDS) | Installation of an NVIS lighting system in a Eurocopter EC 135 (MEGHAS). Major Change to EASA STC 10036349 - in accordance with Master Document List EC 135-02956-001-LMD-00 Rev 0. Note: This revision of the EASA STC 10036349 change introduces a file (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | STC 10036349: (Applicable to EC 135 T2+ EC 135 P2+ EC 135 T2(CPDS) EC 135 P2(CPDS)- Flight Manual Supplement EC 135-01065-025-MFS, Revision OSTC 10036349 Revision 1: (Applicable to EC 135 T2+)- Flight Manual Supplement EC 135-02956-025-MFS, Revision 0o (...) | - The first installation of this kit must be supervised by the STC holder. - This rotorcraft employs Night Vision Imaging System lightning commonly named NVIS lightning, that has been certified to ensure the aircraft is compatible with night vision goggles (...) | 07.10.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|---------------------|----------------|----------------------|---|---|--|--|--|---------------|
| 10039230 | ADAC LUFTFAHRT TECHNIK GmbH | | | | BK117 C-2 | NVIS Certification of mission equipment on an Eurocopter BK117 C2 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS29 Amdt 2 incl. App. B The requirements for environmental protection and the associated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement: EC145-01932-025-MFS-00 Rev. 0 or later revisions of the above listed documents approved by EASA-EC145-01932-023-MMS-01 rev 1 MAINTENANCE MANUAL SUPPLEMENT-EC145-01932-004-DIN-00 Rev. 0 Installation Instructions or later revisions of the above listed documents. | Required pre-modification EASA approval 10038241 (ECD major change 2237) | 19.04.2012 |
| 10044319 | ADAC LUFTFAHRT TECHNIK GmbH | | | | MBB-BK117 B-2, MBB-BK117 C-1 | Installation of an external mirror with electrical remote control as defined in BK117-02441-001-LMD-00. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement BK117-02441-025-MFS Rev. 0 or later revisions of the above listed documents approved by EASA - Master Document List BK117-02441-001-LMD Rev. 0 - Certification Programme BK117-02441-004-RCP Rev. 1, - Installation Instructions B (...) | None. | 10.04.2013 |
| 10044463 | ADAC LUFTFAHRT TECHNIK GmbH | | | | EC 135 P1 (CDS), EC 135 P1 (CPDS), EC 135 P2 (CPDS), EC 135 P2+, EC 135 T1 (CDS), EC 135 T1 (CPDS), EC 135 T2 (CPDS), EC 135 T2+. | Installation of a FLARM | - CS 27 Amdt 2- Original Type Certification Basis Special Condition "Protection of Electrical and Electronic Systems from High Intensity Radiated Fields" (CRI F-7, Issue 5) The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Flight Manual Supplement EC 135-02376-025-MFS-00, Rev 0 or later revisions of the above listed documents approved by EASA - Master Document List EC 135-02376-001-LMD-00, Rev 0 - Maintenance Manual Supplement/ICA EC 135-02376-023-MMS, Rev 0 | None | 15.04.2013 |
| 10028594 | ADVANCED AVIONICS TECHNOLOGIES | P-EASA.IM.A.S.03223 | REV. 1 | FAA STC ST09365SC | DHC-8-201, DHC-8-202 | Installation of dual Collins Mode S Transponders FAA STC ST09365SC, IR, August 14, 1997 | FAR 25 as per EASA.IM.A.191, Issue 4 (incl. SC H-01 ICA on EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Installation of dual Collins Mode S Transponders law. AAT Master Drawing List 1101-000, Rev. B, dated September 18, 2000 - Airplane Flight Manual Supplement Doc. No. 1101-916, Rev. IR, August 14, 1997 - Maintenance law Bombardier Maintenance Manual PSM 1- (...) | None | 19.07.2011 |
| 10028595 | ADVANCED AVIONICS TECHNOLOGIES | P-EASA.IM.A.S.03220 | REV. 2 | FAA STC ST09326SC | DHC-8-201, DHC-8-202 | Installation of Dual Universal Navigation UNS-1C Flight Management System FAA STC ST09326SC, Rev. 1, dated 14 November 2000 | FAR 25 as per EASA.IM.A.191, Issue 4 (incl. SC H-01 ICA on EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Installation of Dual Universal Navigation UNS-1C Flight Management System law AAT Master Drawing List 1061-000, Rev. G - Airplane Flight Manual Supplement Doc. No. 1061-916, Rev. B - Maintenance law Bombardier Airplane Maintenance Manual PSM 1-82-2 or I (...) | None | 20.07.2011 |
| 10028911 | ADVANCED AVIONICS TECHNOLOGIES | P-EASA.IM.A.S.03222 | REV. 2 | FAA STC ST09347SC | DHC-8-201, DHC-8-202 DHC-8-401, DHC-8-402 | Installation of Coltech ATSCALL System FAA STC ST09347SC, Rev. 1, dated May 08, 2001. | FAR Part 25 as per TCDS EASA.IM.A.191, Issue 4 (including Special Conditions H-01 "Enhanced Airworthiness Programme for Aeroplane Systems- ICA on EWIS") for DHC-8-100/200/300 Series CS 25/JAR 25 as per TCDS EASA.IM.A.191 Issue 4 (including Special Cond (...) | - Installation of Coltech ATSCALL law. AAT Master Drawing List 1078-000, Rev. B, dated February 01, 2001 - Airplane Flight Manual Supplement Doc. No. 1078-916, Rev. A, dated May 06, 2001 - Maintenance law Bombardier Airplane Maintenance Manual PSM 1-82-2, (...) | None | 20.07.2011 |
| 10029263 | ADVANCED AVIONICS TECHNOLOGIES | P-EASA.IM.A.S.03221 | REV. 2 | FAA STC ST09346SC | DHC-8-201, DHC-8-202 | Installation of Honeywell Traffic Alert and Collision Avoidance System, TCAS II FAA STC ST09326SC, Rev. 1, dated 14 November 2000. | FAR-25 as per EASA.IM.A.191, Issue 4 (incl. SC H-01 ICA on EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation of Honeywell Traffic Alert and Collision Avoidance System, TCAS II law. AAT Master Drawing List 1065-000, Rev. B, dated March 17, 1997 - Airplane Flight Manual Supplement Doc. No. 1065-916, Rev. IR - Maintenance law Bombardier Airplane Maintenance Manual PSM 1-82-2, (...) | none | 20.07.2011 |
| 10017156 | AERAZUR - ZODIAC AEROSPACE GROUP | EASA.R.S.01263 | REV. 6 | | S-92A | Installation of AERAZUR Internal Auxiliary Fuel System (IAFS) completion kit p/n 508617-1 or 508617-2 or 508617-3. STC revision 1 introduces kit p/n 508617-1 and revised documentation. STC revision 2 introduces edition 15 of RFM supplement ST (...) | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Conditions (SC): IADS CRI C-1 Issue 3 - Crash conditions - Fitting factor and IAFS CRI E-1 Issue 3 - Fuel system crash (...) | S-92A Flight Manual Supplement n° 060S437, edition 16 dated 06 December 2010 or later revisions of the above listed documents approved by EASA: AERAZUR document CBIA n°060S149, revision 11 dated 27 September 2013. AERAZUR document CBIA n°060S147, revisi (...) | The IAFS Completion Kit can be installed only on S-92A helicopter equipped with S-92A Avionics Management System (AMS) version n° 091-2527-007 or later approved version, and AERAZUR IAFS Provision Kit p/n 509429-02 or 509429-03 approved under EASA (...) | 04.10.2013 |
| 10017259 | AERAZUR - ZODIAC AEROSPACE GROUP | EASA.R.S.01466 | REV. 7 | | S-92A | Installation of Provision Kit p/n 509429-02 or p/n 509429-03 of Internal Auxiliary Fuel System (IAFS). STC revisions 1 and 2 introduce minor design modifications for kit p/n 509429-02 and revised documentation. STC revision 3 introduces kit p/n 509429-0 (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Aerazur Organigramme Technique ref. 509429-03 issue 13 dated 05 July 2013; Aerazur Organigramme Technique ref. 509429-02 issue 12 dated 26 May 2010. IAFS - High Profile (HP) version - Maintenance Manual Supplement for provision kit - chapter 11 and 28, iss (...) | Refer to applicable installation instructions for compatibility of the IAFS provision kit with S-92A cabin interior panels as stated in STC 10017259 revisions 1 to 6. Prior to installation of this design change it must be determined that the interrelat (...) | 26.08.2013 |
| 10014306 | AERLYPER S.A. | EASA.A.S.01550 | | | BEECH 200 | Installation of MST67A Transponder Mode S Equipment. | | Master Data List AERLYPER DOC N° 100469 Ed. 04 Instruction for continuous Airworthiness AERLYPER DOC: N° 100482 Ed. 01 | | 06.12.2007 |
| 10014317 | AERLYPER S.A. | EASA.A.S.01567 | | | BEECH 200, SUPER KING AIR | Tactical console installation on B-200 aircraft with ARGOS system, tactical communications system and fixed provisions for ASTRA system. | | Documentation referred to in Doc. No. 100613 "Documents and Drawings List" Issue 04, 27 November 2006. Doc. No. 100528 "Aircraft Flight Manual Supplement", Issue 1, 27 November 2006 included. | In accordance with Doc. No. 100528 "Aircraft Flight Manual Supplement", Issue 01, 27 November 2006. | 31.05.2007 |
| 10014318 | AERLYPER S.A. | EASA.A.S.01568 | | | BEECH 200 | Inmarsat SATCOM installation on board B200. The modification is optional and fixed. It consists of an Inmarsat SATCOM terminal to be installed in the passenger cabin. A SATCOM antenna is installed on the upper part of the fuselage. | | Documentation referred to in: Doc 100423 Ed. 04: Documents and Drawings List. | In accordance with Doc. No. 100429 AFM Supplement ED.03. | 31.05.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|--------------------------------|--------------------------------|--|--|--|--|---------------|
| 10014319 | AERLYPER S.A. | EASA.A.S.01569 | | | BEECH 200 | Installation of New Avionics Equipment to comply with COM/NAV regulations for IFR and B-RNAV Operation. | | Master Data List AERLYPER DOC N° 100542 Ed. 4New Supplemento Aircraft Flight Manual AERLYPER DOC N° 100553 Ed. 00Instructions for Continuous Airworthiness AERLYPER DOC N° 100554Ed. 02 | | 06.12.2007 |
| 10014420 | AERLYPER S.A. | EASA.A.S.01741 | | | FAIRCHILD SWEARINGEN SA 227-BC | Garmin GPS 155XL with antenna GA-56, Mid-Continent MD41-448Annunciation control Unit and copilot CDI Mid-Continent MD 40-09L installation for B-RNAV operation | | Doc No A103-081-01 "Plan de Certification" Edition 04Installation Instructions in accordance withDoc No A103-081-00, "Lista de Documentos y Planos" Edition 05Doc No A103-081-14, "Descripcion de la Instalacion" Edition 01Doc No A103-081-15, "Boletin d (...) | The aircraft is approved for dual pilot B-RNAV operation, subject to the conditions, limitations and procedures contained in Doc NoA103-081-03 "Flight Manual Supplement", Rev 03 | 17.01.2006 |
| 10017103 | AERLYPER S.A. | EASA.R.S.01073 | | | EC 135 P2 | Microwave Transmission System on EC-135 Helicopters. | | Documentation referred in: Doc No. 100189: Documents and Drawings List. | In accordance with Doc No 100198: RFM Supplement Ed. 3 | 14.02.2006 |
| 10017104 | AERLYPER S.A. | EASA.R.S.01075 | | | BO105 C, BO105 S | ASTRA System installation on BO-105 C/S Helicopters | | -Documentation referred in Doc No 100130: Documents and Drawings List | In accordance with Doc No 100137: RFM Supplement Ed.4 | 30.01.2006 |
| 10017105 | AERLYPER S.A. | EASA.R.S.01076 | | | EC 135 P2 | Installation of Image Recorder System UM III and Options System | | Documentation referred in Doc: No 100052: Documents and Drawings List | Doc No 100059: RFM Supplement Ed. 3 | 08.02.2006 |
| 10017137 | AERLYPER S.A. | EASA.R.S.01231 | | | BO 105 C, BO 105 S | Digital microwave transmission system installation on boardBO-105 C/S. | | Doc. No. 100388 "Master Data List", issue 3, June 2008.Doc. No. 100589 "Aircraft Flight Manual Supplement", issue 1, April 2008.Doc. No. 100590 "Continuing Airworthiness Instructions", issue 00, February 2008. | In accordance with Doc. No. 100589 "Aircraft Flight Manual Supplement", issue 1, April 2008.Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously instal (...) | 01.07.2008 |
| 10017197 | AERLYPER S.A. | EASA.R.S.01338 | REV. 1 | | EC-135-P2 (CPDS) EC-135-P2+ | UM-III HD Camera System & Microwave Transmission System Installation Onboard EC135. | | ¿ Master Data List Aerlyper Doc No. 100961 Ed. 03.¿ Supplement to Rotorcraft Flight Manual (RFM) Aerlyper Doc No.100973 Ed. 01.¿ Instructions for Continuous Airworthiness Aerlyper Doc No. 100974 Ed. 01. | 1. Master Data List and RFM Supplement mentioned above. | 13.05.2008 |
| 10033659 | AERLYPER S.A. | P-EASA.R.S.01507 | REV. 1 | | EC135 P2 & EC135 P2+ | Installation of STAR SAFIRE HD Camera System & Digital TransmissionSystem | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalJAR 27 Amendment 3, dated 01/04/2002, for paragraphs 27.561 & 27.602.JAR 27 Amendment 2, (...) | - Document: «Lista de Documentos y Planos», n°: 101499, Edition: 03.Date: 28-01-2011- Document: «Suplemento al Manual de Vuelo», n°: 101511, Edition: 00.Date: 28-01-2011- Document: «Instrucciones de Aeronavegabilidad Continuada», n°: 101512, Edition: (...) | None | 02.03.2011 |
| 10015399 | AERO ASSOCIATES INC | EASA.IM.A.S.01109 | | FAA STC SA00548AT | CESSNA 172 SERIES | Camera Hole Installation | | Aero Associates Inc Drawing ED94003 no revision dated march25, 1994 or later FAA approved revision as defined in FAA STC SA00548AT | 1. Continued Airworthiness-All new and modified structure that forms part of this STC must be inspected at least annually for damage. Any defects found must be repaired by approved repair scheme.2. This STC is approved only for the productconfigurat (...) | 30.05.2005 |
| 10015649 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.01624 | REV 1 | TCCA STC C-LSA05-303/D ISSUE 2 | BOMBARDIER CL-600-2B16 | Aircraft Completion 12 Passenger Interior Configuration | | - Aero Consulting Services Modification Data Summary E596000 Revision B or later approved revision.- Midcoast Aviation AFM Supplement 34-424M005 Revision N/C or later approved revision.- Midcoast Aviation AFM Supplement 34-261M017Revision N/C or later approved revision. | Restricted to Aircraft Serial Number 5610 | 25.07.2006 |
| 10015653 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.01628 | REV 1 | TRANSPORT CANADA STC C-LSA05-2 | BOMBARDIER CL-600-2B16 | Aircraft Completion 12 Passenger Interior Configuration | | - Aero Consulting Services Modification Data Summary E281000 Revision B or later approved revision.- Midcoast Aviation AFM Supplement 34-424M005 Revision N/C and 34-261M017 Revision N/C or later approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer.2. Restricted to Aircraft Serial Number 5609. | 25.07.2006 |
| 10015670 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.01647 | REV 1 | TCCA STC C-LSA05-317/D ISSUE | BOMBARDIER CL-600-2B16 | Aircraft Completion 12 Passenger Interior Configuration | | -Transport Canada STC C-LSA05-317/D Issue 2-Aero ConsultingServices Modification Data Summary E611000 Revision A or later approved revision-Midcoast Aviation AFM Supplement 34-261M017 Revision N/C & 34-424M005 Revision N/C or later approved revision | 1. Restricted to ASN 5620 only.2. This STC is approved onlyfor the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 10.08.2006 |
| 10015671 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.01649 | REV 1 | TCCA C-LSA05-315/D ISSUE 2 | BOMBARDIER CL-600-2B16 (CL-604 | Aircraft Completion 12 Passenger Interior Configuration | | -Transport Canada STC C-LSA05-315/D Issue 2-Aeroconsulting Services Modification Data Summary E613000 Revision A or later approved revision-Midcoast Aviation AFM Supplement 34-261M017 Revision N/C & 34-42M005 Revision N/C or later approvedrevision | 1. Restricted to ASN 5612 only.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.08.2006 |
| 10015678 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.01659 | REV 1 | TCCA STC C-LSA05-328/D | BOMBARDIER CL-600-2B16 | Aircraft Completion 12 Passenger Interior Configuration | | Transport Canada Limited STC C-LSA05-328/D Issue 2 dated October 02 2006.Aero Consulting Services Modification Data Summary E640000 Revision A or later approved revisionMidcoast Aviation Airplane Flight Supplement 34-424M005 Revision N/C and 34-261M017 Revision N/C or later approved revision | 1. Limited to S/N 56172. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 27.10.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|--------------------------------|---|---|---------------------|---|---|---------------|
| 10015859 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.01967 | | TCCA STC SA03-43 | BOMBARDIER CL-600-2B19 | Installation of Safe Flight Autopower (Autothrottle) System | | Transport Canada STC SA03-43 Issue 1 dated May 09, 2003AeroConsulting Services Modification Data Summary D45150Safe Flight AFM Supplement R-2827 Revision N/C dated May 09 2003 or later approved revision | -- | 14.09.2006 |
| 10016132 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.02326 | | TCCA STC SA95-67 | BOMBARDIER DHC-8-100 SERIES BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-300 SERIES | Installation of Aero Consulting Services Ltd. "Improved Exhaust Outlet Shroud Assembly" according to Transport Canada STC SA 95-67 | | The manufacture and installation of the Improved Exhaust Outlet Shroud Assembly is to be completed in accordance with Aero Consulting Services Ltd. Modification Data Summary Number A23820, Revision D, dated March 27, 2001 or later DOT approved revis (...) | | 01.03.2007 |
| 10016192 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.02429 | | TCCA SA01-65, ISSUE 4 | BOMBARDIER DHC-8-100 SERIES BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-300 SERIES | Addition of filtration to Cabin Recirculation Air. Addition of recirculation air filter reduces potential blockage of the condenser mixer and promotes peak performance of the ECS system. TCCA STC SA01-65, Issue 4, dated 24 June 2005. | | Aero Consulting Services Modification Data Summary No. C46000 Revision D or later approved revisions for DHC-8-100 and DHC-8-200 series or S46030 Revision C or later TCCA approved revisions for DHC-8-300 series. Maintenance Manual Supplement (MMS) C (...) | 1. Applicable to DHC-8-100 series with 1990 (Post-Mod 8/1127*) cabin interior, all DHC-8-200 series and DHC-8-300 series with 1990 (Post-Mod 8/1128*) cabin interior.*Including Hunting and CD interior.2. Prior to installation of this modificali (...) | 06.07.2007 |
| 10016270 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.02554 | | TCCA STC C-LSA07-190/D ISSUE 2 | | Upgrade to Honeywell Mark V EGPWS. | | Aero Consulting Services Modification Data Summary F37900, rev B Flight Manual Supplement, F379091, rev N/C. Instructions for Continued Airworthiness, F379071, rev N/C. | | 02.11.2007 |
| 10016376 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.02699 | REV. 1 | TCCA LSTC C-LSA07-342/D | BOMBARDIER BD-700-1A11 | Installation, Side Facing Divan, S/N 9229 Configuration. | | Transport Canada Statement of Compliance C-07-0946 dated 22January 2008. Transport Canada LSTC No. LS707-342/D Issue 1 dated 06 December 2007 Aero Consulting Services Ltd. Modification Data Summary No. F545000 Revision N/C or later approved revision. (...) | Restricted to ASN 9229. | 05.03.2008 |
| 10016429 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.02768 | | TCCA STC C-LSA08-017/D ISSUE 2 | BOMBARDIER BD-700-1A10 | Cabin Electronics Upgrade. | | Transport Canada Limited STC No. C-LSA08-017/D dated 06 February 2008Transport Canada Statement of Compliance ref C-08-0114 dated 04 July 2008Aero Consulting Services ModificationData Summary No. F60200 Revision NC or later approved revision | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 09.07.2008 |
| 10016470 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.02816 | | TCCA STC C-LSA08-143/D | GULFSTREAM IV | Installation of Interior Modifications. | | Installation in accordance with Aero Consulting Services Ltd Modification Data Summary F610000 Rev N/C or later approved revisions. Maintenance in accordance with Aero Consulting Services Ltd Mandatory Maintenance Manual Supplement F610071 Rev N/C an (...) | Aircraft Serial Number 1109 Only.Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverseeffect upon the airworthiness of the product. | 02.06.2008 |
| 10016471 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.02817 | | TCCA STC C-LSA08-151/D | GULFSTREAM IV | Installation of Rockwell Collins Radio Tuning Units RTU-4210 and Rockwell Collins TDR-94D (ELS/EHS) Transponder. | | Installation in accordance with Aero Consulting Services Ltd. Modification Data Summary F77300 Rev N/C or later approved revisions. Maintenance in accordance with Aero Consulting Services Ltd Mandatory Maintenance Manual Supplement F773073 Rev N/C, a (...) | Application to Aircraft Serial Number 1109 only. The operator is responsible for th inclusion of a 24 month check in their maintenance schedule of the Mode S transponder system to ensure correct operation of the 24BIT address and the required DAPs IA (...) | 11.06.2008 |
| 10016605 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.03022 | | TCCA STC SA00-109, ISSUE 4 | BOMBARDIER DHC-8-102 BOMBARDIER DHC-8-103 BOMBARDIER DHC-8-200 SERIES | Installation of Environmental Control System, Low Temperature Limit System (Cold Clamp)Ref. TCCA STC SA00-109, Issue 4, dated 19. October 2000 | | Modification has to be completed in accordance with ACS Modification Data Summary, C 46400, Rev. C, dated 05. July 2008 "ACS Maintenance Manual Supplement, C46481, Rev. A, dated 15. January 2007" or later revisions of the above listed documents approv (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness ofthe pr (...) | 19.12.2008 |
| 10016606 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.03023 | | TCCA STC SA02-4, ISSUE 1 | | Installation of NAI part C93602 Solid State Logo Light | | Transport Canada STC No. SA02-4 Issue 1 dated 9 January 2002.Transport Canada SOC Reference C-07-0114 dated 9 October 2008.Aero Consulting Services Modification Data Summary No. C93600 Revision B or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 14.10.2008 |
| 10016612 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.03033 | | TCCA STC C-LSA08-268/D | BOMBARDIER BD-700-1A10 | Installation, Side Facing Divan, S/N 9260 Configuration | | TCCA STC No. C-LSA08-268/D Issue 1 dated 18 September 2008.TCCA SoC ref. C-08-0759 dated 28 November 2008. Aero Consulting Services Ltd. Modification Data Summary Number F957000 Revision N/C or later approved revision.Aero Consulting Services Ltd. AI (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 13.03.2009 |
| 10016629 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.03064 | | TCCA STC C-LSA09-087/D | GULFSTREAM IV | Refurbishment and Installation of Cabin Interior Modifications. | | Installation in accordance with Aero Consulting Services Ltd Modification Data Summary G040000 Rev N/C or later approved revisions. Maintenance in accordance with Aero Consulting Services Ltd Instructions for Continued Airworthiness G040071Rev NC or later approved revisions. | Aircraft Serial Number 1202 Only.Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 08.04.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------------|----------------|--------------------------------|---|--|---|---|--|---------------|
| 10016630 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.03065 | | TCCA STC C-LSA09-086/D | GULFSTREAM IV | Installation of Rockwell Collins Radio Tuning Units RTU-4210 and Rockwell Collins TDR-94D (ELS/EHS) Transponder. | | Installation in accordance with Aero Consulting Services Ltd Modification Data Summary G039000 Rev N/C or later approved revisions.Maintenance in accordance with Aero Consulting Services Ltd Instructions for Continued Airworthiness F773071Rev A and (...) | Aircraft Serial Number 1202 OnlyThe operator is responsible for the inclusion of a 24 month check in their maintenance schedule of the Mode S transponder system to ensure correct operation of the 24BIT address and the required DAPs IAW ACS MM 34 53 00. | 08.04.2009 |
| 10016647 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.03091 | | TCCA C-LSA09-088/D | | Installation of L3 Communications MicroQAR Quick Access Recorder. | | Installation in accordance with Aero Consulting Services Ltd Modification Data Summary G052000 Rev N/C or later approved revisions.Maintenance in accordance with Aero Consulting Services Ltd Instructions for Continued Airworthiness G052071Rev NC or later approved revisions. | Aircraft Serial Number 1202 OnlyPrior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 08.04.2009 |
| 10016662 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.03125 | | TCCA LSTC # C-LSA08-317/D ISSU | BOMBARDIER CL-600-2B16 | Installation, Complete Custom Aircraft Interior | | TCCA STC No.C-LSA08-317/D Issue 1 dated 12 December 2008.Aero Consulting Services Ltd. Modification Data Summary G048000 Revision NC or later approved revision.Midcoast Aviation Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 19.02.2009 |
| 10016664 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.03129 | | TCCA STC NO. C-LSA09-022/D ISS | BOMBARDIER CL-600-2B16 | Installation, Complete Custom Aircraft Interior. | | TCCA STC No. C-LSA09-022/D Issue 1 dated 03 April 2009. - Aero Consulting Services Ltd. Modification Data Summary G101000 Revision NC* - Midcoast Aviation Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Inst (...) | 1. Restricted to ASN 5734.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the (...) | 14.04.2009 |
| 10016674 | AERO CONSULTING SERVICES LTD. | EASA.IM.A.S.03177 | REV. 1 | TCCA STC SA91-97 | BOMBARDIER DHC-8-100 SERIES BOMBARDIER DHC-8-103 BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-300 SERIES | Safety Catch Mechanism on the Rear Baggage DoorRef. TCCA STC SA91-97, Issue 3, dated 18.11.2003Note: This STC supersedes EASA.IM.A.S.03177. | | Installation in accordance to ACS TCCA approved Modification Data Summary Report 956900, Rev. E, dated 28.06.2007*).Maintenance in accordance with ACS Maintenance Manual Supplement, 956971, Rev. B, dated 08 November 2006*) or later revisions (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 14.05.2009 |
| 10027216 | AERO CONSULTING SERVICES LTD. | P- EASA.IM.A.S.03160 | | TCCA STC C-LSA09-079/D | BD-700-1A11 | Installation 16G Side Facing Divan | EASA TCDS No IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approvalThe certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | -TCCA STC No C-LSA09-079/D Issue 2 dated 14 August 2009.-TCCA SoC ref. C-09-0745 dated 24 August 2009.-Aero Consulting Services Ltd. Modification Data Summary Number G198000Revision A.-Aero Consulting Services Ltd. Airplane Flight Manual Supplement (...) | Restricted to: ASN 9289 | 14.09.2009 |
| 10027715 | AERO CONSULTING SERVICES LTD. | | | TCCA LSTC C-LSA09-099/D | BD-700-1A10 | Installation, 16 G Side Facing Divan. | EASA TCDS No IM.A.009.The Certification Basis for the original product remains applicable to this certificate.The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate. | TCCA STC No. C-LSA09-099/D Issue 1 dated 14 October 2009.TCCA SoC ref. C-09-0969 dated 21 October 2009.Aero Consulting Services Ltd. Modification Data Summary Number G255000Revision N/C or later approved revision.Aero Consulting Services Ltd. Airpla (...) | None. | 26.10.2009 |
| 10028465 | AERO CONSULTING SERVICES LTD. | | | TCCA LSTC C-LSA08-007/D | LEARJET 60 | Installation of Chelton HSD Satcom System. | The Certification Basis for the original product remains applicable to this certificate/ approvalThe certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | F616000 Rev D - Modification Data SummaryF616071 Rev B - Instructions for Continued AirworthinessF616072 Rev A - Mandatory Maintenance Manual Supplementor later revisions of the above listed documents approved by EASA in accordance with EASA ED Decis (...) | Approval is valid for s/n 60-328 only. | 13.01.2010 |
| 10028799 | AERO CONSULTING SERVICES LTD. | | | TCCA STC C-LSA10-006/D | 900XP | Installation of Dual Lifeport Patient Loading and Utility System (PLUS) | | CAR 4b and FAR 25 Installation in accordance with Aero Consulting Services LtdModification Data Summary G569000 Rev N/C or later approved revisions- Operation in accordance with Aero Consulting Services Ltd FlightManual Supplement G569090 Rev N/C or later approved r (...) | Aircraft Serial Number HA-0114 Only | 10.02.2010 |
| 10029239 | AERO CONSULTING SERVICES LTD. | | | TCCA LSTC C-LSA10-010/D | PC-12/47E | Installation, Lifeport PLUS Units in Tandem Configuration | Canadian Airworthiness Manual Chapter 523, Change 3 dated January 2, 1992, which is equivalent to 14 CFR 23, Normal Category, effectiveFebruary 4, 1991, including Amendments 23-1 through 23-42.The Certification Basis for the original productand the fo (...) | Installation in accordance with ACS-NAI Ltd. Modification Data SummaryG456000 Rev N/C or later approved revisions.-Operation in accordance with Pilot's Operating Handbook andTCCA-Approved Airplane Flight Manual Supplement EGR-855-11-02 Rev A *PLUS (...) | The applicability of the present STC is limited to AircraftSN: 1165Refer to the Limitations section in the TCCA-approved Airplane FlightManual Supplement listed in associated technical documentation | 15.03.2010 |
| 10029773 | AERO CONSULTING SERVICES LTD. | | REV. 1 | TCCA PROJECT # C-09-784 | PC-12/47E | Main battery No. 2 relocation.Validation of the TCCA STC SA10-4 Issue 1: Relocation of the mainbattery No 2 from the tail section to the engine compartment. | FAR 23The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificatednoise and/ or emissions levels of the original product areunchanged and (...) | - Modification Data Summary, Doc. No. G409000 Rev. C- Pilot's Operating Handbook Supplement, Doc. No. G409091 Rev. N/C- ICA- Maintenance Manual Supplement, Doc. No. G409071 Rev. B & IPC, Doc. No. G409072 Rev. B, or later revisions of the above listed doc (...) | None. | 02.11.2010 |
| 10029833 | AERO CONSULTING SERVICES LTD. | | | TCCA STC SA10-9 | CL600-1A11, CL600-2A12, CL600-2B16 | Transport Canada STC No. SA10-9 Issue 1 dated 08 February 2010.ACS-NAI Ltd. Modification Data Summary Number G373000 Revision N/C or later approved revision:Instructions for Continued Airworthiness G373071 Revision A and G373072Revision N/C, or later (...) | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approvalThe certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | | None | 23.04.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|----------------------|----------------|----------------------------|----------------------------|---|--|---|---|---------------|
| 10030100 | AERO CONSULTING SERVICES LTD. | | | TCCA STC NO. C-LSA10-048/D | BD-700-1A10 | Installation, Complete Custom Aircraft Interior | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval | TCCA STC No. C-LSA10-048/D Issue 1 dated 07 May 2010.TCCA SoC ref. C-10-0416 dated 19 May 2010.ACS-NAI Ltd. Modification Data Summary G568000 Revision NC or later approved revision.Midcoast Aviation Inc. Airplane Flight Manual Supplement for the Compl (...) | Restricted to ASN 9319 | 21.05.2010 |
| 10030139 | AERO CONSULTING SERVICES LTD. | | | TCCA STC NO. C-LSA10-041/D | BD-700-1A11 | Installation, Complete Custom Aircraft Interior | EASA.IM.A.009The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.CRI F-01 "Installation of Wireless System" The certificated noise and/ or em (...) | TCCA STC No. C-LSA10-041/D Issue 1 dated 07 May 2010.TCCA SoC ref. C-10-0427 dated 21 May 2010.ACS-NAI Ltd. Modification Data Summary G363000 Revision NC.Savannah Air Center Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior, (...) | Restricted to ASN 9324 | 27.05.2010 |
| 10016126 | AERO DESIGN LTD. | EASA.IM.A.S.02320 | | TCCA STC C-LSA05-204/D | PIPER PA-31 (NAVAJO) | Installation of Heater Exhaust Extension. / The purpose is to route exhaust from the combustion heater in the aircraft nose to the aft of the wing to prevent fouling of the camera port. Ref.: Transport Canada Limited Supplemental Type Certificate C-LSA05-204/D Issue 2. | | Aero Design Ltd. Document Control List DLC637 Rev. 0, dated 6 April 2005, Listing Engineering Documents, Fabrication Documents and Installation Documents. Aero Design Ltd. Engineering Report ER637.01.01.Air Worthiness Compliance Programme. | Limited to aircraft serial numbers 31-457 and 31-7912028. | 28.08.2007 |
| 10016782 | AERO DESIGN LTD. | EASA.IM.R.S.01123 | | TCCA STC SH02-26, ISSUE 2 | EUROCOPTER FRANCE AS 350 B | Installation of DynaNav Seisbag System. | | Definition and installation: Aero Design Ltd. Document Control List DCL444, Revision 2, or later Transport Canada (TCCA) approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 25.01.2008 |
| 10043360 | AERO DESIGN LTD. | | REV. 1 | TCCA STC SH12-58 ISSUE 2 | 429 | INSTALLATION OF EXTERNAL ATTACHMENT PROVISIONS, CARGO BASKET, AND CABIN STEPS. Configuration A - External Attachment Provisions described in AERO Design Ltd. Document Control List, DCL959-2, Revision 0, dated 28 September 2012, or later approved revision (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Configuration A and B: AERO Design Ltd. Flight Manual Supplement FMS959.90, Revision 0, dated 08 November 2012 Configuration A and B: AERO Design Ltd. Instructions for Continued Airworthiness, ICA959.91, Revision 0, dated 28 September 2012 Configuration (...) | External Attachment Provisions installed in accordance with DCL959-2 may remain installed in accordance with DCL959-2 if the basket installation is removed. Installation of Configuration A, External Attachment Provisions, is a prerequisite for installation of Configuration B (-> ->) External Cargo Basket T | 25.01.2013 |
| 10028376 | AERO DYNAMIX, INC. | P- EASA.IM.R.S.01543 | | FAA STC SR09480RC | MD900 | Installation of Night Vision System (NVIS) / Night Vision Goggles (NVG) Cockpit lighting. Validation of FAA STC SR09480RC. | JAR27 1st amendment The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | -Aero Dynamix, Inc. Master drawing list MDL 1775-01, Revision M, dated 21 September 2009, or later approved revisions ("") Aero Dynamix, Inc. Instructions for continued airworthiness (ICA), document No. ICA MD900-01, Revision N, dated 22 December 2009, (...) | Refer to applicable RFMS for list of equipment mandatory for Night Vision Image System certification | 23.12.2009 |
| 10031298 | AERO DYNAMIX, INC. | | | FAA STC SR09519RC | AW119MKII | Installation of Night Vision System (NVIS) / Night Vision Goggles (NVG) Cockpit lighting. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Rotorcraft Flight Manual Supplement (RFMS) document RFSAW119MKII-01-EASA revision -, dated 05 August 2010, or later revisions of the above listed documents approved by EASA - Instructions for Continued Airworthiness (ICA), document Aero Dynamix, Inc. (...) | Refer to applicable RFMS for list of equipment mandatory for Night Vision Image System certification The Finish Board Guard Configuration as defined in Appendix B of document "Instructions for Continued Airworthiness (ICA)", Aero Dynamix, Inc. No. ICA A1 (...) | 13.08.2010 |
| 10034881 | AERO DYNAMIX, INC. | | | FAA STC SR09543RC | S-76C | Installation of Night Vision Imaging System (NVIS) / Night Vision Goggle (NVG) compatible lighting system. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Master Drawing List (MDL), Aero Dynamix, Inc. drawing number MDL4783-01, revision D, dated 8 December 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this (...) | Refer to applicable Rotorcraft Flight Manual Supplement (RFMS) for list of equipment mandatory for Night Vision Imaging System (NVIS) certification. | 10.05.2011 |
| 10042804 | AERO DYNAMIX, INC. | | | FAA STC SR09566RC | 429 | Installation of Night Vision Imaging System (NVIS) / Night Vision Goggle (NVG) compatible lighting system. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Master Drawing List (MDL), Aero Dynamix, Inc. drawing number MDL5600-01 revision J, dated 12 November 2012; Rotorcraft Flight Manual Supplement (RFMS), document number RFSM429-02, revision IR, dated 13 June 2012; or later revisions of the above list (...) | Refer to applicable Rotorcraft Flight Manual Supplement (RFMS) for list of equipment mandatory for Night Vision Imaging System (NVIS) certification. | 19.12.2012 |
| 10014080 | AERO ENGINEERING DESIGN LIMITED | EASA.A.S.01172 | | | BOEING 737-300-400-500 | Aero Engineering Design Modification 34-1-226.SSR Mode S Enhanced Surveillance | | Aero Engineering Design Modification Certificate 34-13-226 Issue 1 dated May 9 2005 or later approved revision | | 21.06.2005 |
| 10014081 | AERO ENGINEERING DESIGN LIMITED | EASA.A.S.01173 | | | BAE HS 748 | Installation of a Honeywell MK VI EGPWS | | -AED Design Modification Certificate 34-22-225 dated February 10, 2005 or later approved revision - Aero Engineering Design Flight Manual Supplement AED/748/005 Issue 1 or later approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 02.03.2005 |
| 10014130 | AERO ENGINEERING DESIGN LIMITED | EASA.A.S.01240 | | | BOEING 727-200 | Installation of Honeywell MK VII EGPWS. | | Design Modification Certificate 34-12-237 Issue 1 11/04/2005. AFM-AED/B727/001. | | 22.04.2005 |
| 10043011 | AERO ENGINEERING DESIGN LIMITED | EASA.A.S.01123 | | | 737-200 | Introduction of Honeywell Mk VII EGPWS | | -Design modification certificate 34-13-224 Issue 1 dated February 23, 2005 or later EASA approved revision - Aero Engineering Design Flight Manual Supplement AED/B737/003 Issue 2 February 2005 or later EASA approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 09.03.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|-------------------|----------------|-----------------------|--|--|--|--|--|---------------|
| 10043102 | AERO ENGINEERING DESIGN LIMITED | EASA.A.S.01034 | | | A330 | Introduction of FLIGHTVU Camera Surveillance System | | -25-71-426 Rev 1 dated 25 February 2005 | 1.This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 04.03.2005 |
| 10014186 | AERO ENGINEERING DESIGN Ltd | EASA.A.S.01324 | | | BAE 125 800B | Introduction of Rockwell Collins TCAS 4000 | | Flight Manual Supplement S.21 FMS.125.0004 | | 19.09.2005 |
| 10014187 | AERO ENGINEERING DESIGN Ltd | EASA.A.S.01325 | | | MD-80 | Aero Engineering Design Modification 34-77-211. Introduction of Honeywell MK VII EGPWS | | -Aero Engineering Design Ltd Design Modification Certificate 34-77-211-Aero Engineering Design Ltd AFM Supplement AED/MD80/001 | | 25.07.2005 |
| 10014188 | AERO ENGINEERING DESIGN Ltd | EASA.A.S.01328 | | | MD-80 | Aero Engineering Design Modification 34-77-210. Introduction of Honeywell RDR-4B Weather Radar | | Aero Engineering Design Ltd Design Modification Certificate 34-77-210 | | 25.07.2005 |
| 10016210 | AERO MEDICAL PRODUCTS Co. Inc. | EASA.IM.A.S.02456 | | FAA STC SA1035GL | PIPER PA-31 PIPER PA-31-300 PIPER PA-31-325 PIPER PA-31-350 | FAA STC SA 1035GL "Aircraft Stretcher System - Installation of Air Ambulance Stretcher P/N 125 Kit P/N 800, Medical Oxygen System Kit P/N 807 (FAA approved option), Inverter/ Pump Kit P/N 812 in accordance with Drawing List Report 001 Rev.C or later approved revision. | | - AFM Supplement EA Form FMS FAA Approved 09-02-2004 or later approved revision.- AMM Supplement EA Forms MMS FAA Approved 10-02-2004 or later approved revision.- Installation and Operating Instructions EA Form 109P FAA approved 23-10-2001 o (...) | 1. Prior to the first flight of each modified aircraft, and after each exchange to any of the Communication, Navigational or MedPack Equipment, a ground EMI test must be conducted in accordance with EA Form 173 dated 15-11-1993. A copy of the EA form 173 must be kept with the aircraft records. | 01.02.2008 |
| 10029339 | AERO MEDICAL PRODUCTS Co. Inc. | | | FAA STC ST01188CH | CESSNA 501, 551 | Installation of Air Ambulance Conversion law. Aero Medical Products Mfg Installation Drawing List Report 006-1 or later FAA approved revisions | CS-23.B The Certification Basis for the original product remains applicable to this certificate/ approval | Installation Drawing List Report 006-1 Airplane Flight Manual Supplement to ST01188CH (EA Form 171-C)MMS to ST01188CH or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | As referred in limitations section of the AFM-SPrior to the first flight of each modified aircraft, and after each exchange to any of the communication, Navigational, or Med Pack equipment, a ground EMI test must be conducted in accordance with EA Form 173. A copy of EA Form 173 must be kept with the aircraft records | 19.03.2010 |
| 10034128 | AERO MEDICAL PRODUCTS Co. Inc. | | | FAA STC ST02863CH | FALCON 50 | Aircraft Medical stretcher for MF 50 aircraft FAA STC ST02863CH | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Aeromedical Products doc. "Compliance Check List Falcon 50 - project number ST744CH-T" Revision 6 dated 02/03/2011 Aeromedical Products Mfg., Inc. Drawing List Report 007 Rev 0 dated 22/12/2010. FAA approved Airplane Flight Manual Supplement dated 28.06 (...) | This STC does not constitute an approval for installation. Compliance with minimum following requirements below need to be demonstrated through a design approval for installation in each aircraft model to be equipped with this medical stretcher assembly (...) | 09.03.2011 |
| 10044690 | AERO MEDICAL PRODUCTS Co. Inc. | | | FAA STC SA1130GL | CESSNA 441 | Installation of EMS stretcher system using installation kit 801, including oxygen system 807, inverters and pump kit 812 in accordance with FAA STC SA1130GL. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation and Operating Instructions, Form EA 109K, dated April 23, 2001 Installation and Operating Instructions, Form EA 170, dated June 9, 2001 Aircraft Flight Manual Supplement dated April 18, 2005 Aircraft Maintenance Manual Supplement, Form 107, Rev 2 dated January 15, 2013 or later revisions of t | Prior to the first flight of each modified aircraft, and after each exchange to any of the Communication, Navigational, or Med Pack equipment, a ground EMI test must be conducted in accordance to EA Form 173, dated November 15, 1993 or later FAA approved (...) | 29.04.2013 |
| 10017055 | AERO PACIFIC | EASA.IM.R.S.01486 | REV. 1 | FAA STC STC SR01521LA | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 D | Installation of a Belly Utility Mount. Note: This STC supersedes EASA.IM.R.S.01486. | | - Aero Pacific Master Data List Doc. No. 350UM02 30, Rev. 1 dated 24 January 2006 or later EASA approved revisions.- Aero Pacific Rotorcraft Flight Manual Supplement Doc. No. 350UM02 10, initial issue, dated 16 July 2003 or later EASA approved revisions. | 1. This STC provides approval for structural provisions for installation of other equipment. This other equipment for which these provisions are intended has not been certified. A separate EASA approval is required for the installation of this eq (...) | 04.03.2009 |
| 10039781 | AERO PACIFIC | | | FAA STC SR01688LA | 206, 206A, 206B, 206B-1, 206L 206L-1, 206L-3, 206L-4, 407 | Installation of Aero Pacific Belly Utility Mount AP206UM02 (Validation of FAA STC SR01688LA) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | RFM Supplement N°AP206-UM02-10, issued March 22, 2005 or later revisions of the above listed documents approved by EASA- „Master Drawing List N°20602, Revision A, dated February 1, 2005- „AML N° SR01688LA- „Instructions for Continued Airworthiness (...) | - This Belly utility Mount Installation is approved for a maximum weight of not more than 150 pounds and a frontal area of 2.5 square feet. - This STC only addresses structural, performance and handling qualities requirements. No system or electrical r (...) | 23.05.2012 |
| 10016841 | AERO SEATS AND SYSTEMS, INC | EASA.IM.R.S.01214 | | SR09455RC | SIKORSKY S-76A | Installation of External Life raft Deployment System | | CHC Scotia Report Number CHCE-2500-006 mod asi 500-100 01/02 Revision - dated January 18, 2006 or later approved revision.RFM Supplement S-76-500-FMS Revision IR dated March 1, 2006 or later approved revision. | Approval limited to Sikorsky S-76A registration G-BVCX Serial number 760183 | 11.05.2006 |
| 10016842 | AERO SEATS AND SYSTEMS, INC | EASA.IM.R.S.01218 | | SR09355RC | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of Passenger Seats | | ASI-6409 - Installation Instructions | None | 10.05.2006 |
| 10016844 | AERO SEATS AND SYSTEMS, INC | EASA.IM.R.S.01220 | | FAA STC SR09011RC | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of Emergency Exit Window in Passenger Door | | ASI-7611-MDL dated 5/8/1995 Master Drawing List ASI-95-002 rev A dated 5/8/1995 Evacuation Demonstration report ASI-7611-SI rev A dated 5/9/1995 Service Instruction | Configurations with forward facing seats only | 27.07.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|---------------------|----------------|----------------------|--|--|---|---|---|---------------|
| 10016912 | AERO SEATS AND SYSTEMS, INC | EASA.IM.R.S.01293 | | FAA STC SR09211RC | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of External Mounted Life Raft Deployment System (Identified as #Basic STC# by the FAA and approved using the FAA/JAA/EASA type validation principles (TVP)) | | ASI-500-100 Revision E dated 4/14/1999 Installation manual S76-501-FMS Revision IR dated 1/11/2007 Flight Manual Supplement S-76-500-01 Revision 3 dated 12/21/2006 Instructions for Continued Airworthiness | Installation of external mounted life raft deployment system, with optional deployment handles in passenger compartment, in accordance with Aero Seats and Systems Inc. Installation manual ASI-500-100-01/02, revision H, dated December 5, 2005 or later approved revision is required. | 15.01.2007 |
| 10016924 | AERO SEATS AND SYSTEMS, INC | EASA.IM.R.S.01309 | | FAA STC SR09453RC | BELL 412 BELL 412 EP | Installation of Cabin Seats | | Aero Seats and Systems, Inc., Master Installation List L412-2520-002, Revision E, dated May 9, 2006, or later approved revision. Rotorcraft Flight Manual Supplement, ASI-6409-412-300, Revision A, dated May 18, 2006 or later approved revision/Instructi (...) | | 15.03.2007 |
| 10014192 | AERO TECHNICS DESIGN LIMITED | EASA.A.S.01334 | | | 1. BOEING 747-100 1. BOEING 747-200 1. BOEING 747-400 2. BOEING MD-11 SERIES 3. DC-10 SERIES | Introduction of Horsebox Containers BAC1136-301 and BAC1136-401 | | ATD0061 | | 26.07.2005 |
| 10014600 | AERO TECHNICS DESIGN LIMITED | EASA.A.S.02722 | REV 1 | | BOEING MD-83 | Cabin Reconfiguration from 167Y to 52C/22Y Seating | | Aero Technics Design Modification ATD0150, Aero Technics Design AFM Supplement ATDFMS0150 | | 28.09.2006 |
| 10029250 | AERO TECHNICS DESIGN LIMITED | | | | A300, A310 B747-400, DC-8, DC-10, MD-11, B767, L100L1011, B777 | Introduction of Instone Airservices Airstable Horse/Freight Container P/N 73-90-343. | Various as applicable to aircraft types listed for installation. The Certification Basis for the original product remains applicable to this certificate/ approval. | Certification Plan ATD0311 or later revisions of the above listed documents approved by EASA | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph #Description#. Compatibility with all aircraft configurations shall be determined by the installer. | 26.03.2010 |
| 10031612 | AERO TECHNICS DESIGN LIMITED | | | | B737-500 | Cabin Seating Reconfiguration from 120Y to 16C/28C layout (44C) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ATDDR0382, Issue 1 - Design Report dated August-2010 ATDCCLO382, Issue 6 - Compliance Check List dated August-2010 or later revisions of the above listed documents approved by EASA | Limited to B737-500 S/N 24645, 24648 and 24649 | 31.08.2010 |
| 10031614 | AERO TWIN, Inc. | | | FAA STC SA4625NM | CESSNA 208, 208B | Fabrication and Installation of Aero Twin, Inc EB8-200 extended baggage kit in accordance to FAA STC SA4625NM | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Airplane Flight Manual Supplement EB8-200-AFMS, Rev. B dated October 14, 2004. Instructions for Continued Airworthiness EB8-200-ICA, last revision dated October 14, 2004. Master Drawing List MDL EB8-200, last revision dated March 6, 2009 or later revisions o (...) | None | 31.08.2010 |
| 10033295 | AERO TWIN, Inc. | P-EASA.IM.A.S.02690 | | FAA STC SA02291AK | CESSNA 208, 208B | Installation of Honeywell TPE331-12JR turbine engine and Hartzell HC-B4TN-5CL/LT10890N(K) propeller and associated structure and equipment in accordance with FAA STC SA02291AK. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: 14 CFR Part 23 effective February 1, 1965, as amended by Amendments 23-1 through 23-28 A (...) | Aero Twin, Inc. Master Descriptive Data List 850-MDDL, issue 6, dated January 1, 2009. Cessna 208 (208A), Airplane Flight Manual Supplement, Doc. No. 850-AFMS, Rev. B, FAA approved May 14, 2010. For Cessna 208B, FAA approved Airplane Flight Manual Sup (...) | a) This modification is not compatible with Cessna 208 serial numbers 20800001 through 20800013, 20800416, 208000500 and greater. ii. Cessna 208B serial numbers 208B1190, 208B1216, 208B2000 and greater. iii. FAA STC SA01478W(b) This modification has bee (...) | 12.01.2011 |
| 10037001 | AERO TWIN, Inc. | | | FAA STC SA4626NM | CESSNA 208, 208B | Fabrication and Installation of Auxiliary dry air pump kit AP8-100 in accordance to FAA STC SA4626NM. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Airplane Flight Manual Supplement, Rev. B, dated November 27, 2006. Instructions for Continued Airworthiness, Doc. No. AP8-ICA, Rev. D dated June 22, 2004. Master Drawing List No. AP8-100, dated October 13, 2006. Installation Instructions, Doc. No. AP8-1M, (...) | None | 24.10.2011 |
| 10038365 | AERO TWIN, Inc. | | | FAA STC SA02080AK-D | 208, 208B | Fabrication and Installation of Aero Twin, Inc. Exhaust Deflector P/NED8-100 in accordance with FAA STC SA02080AK-D | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Aero Twin, Inc. Master Data List ED8-100, dated July 15, 2002 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 09.02.2012 |
| 10027700 | AERO VODOCHODY AEROSPACE A.S. | P-EASA.A.S.03909 | REV. 2 | | EMB-135BJ | EFB class 2 navAero-1.Bag C2 ² Electronic Bag System implementation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: Cert Basis CS-25 AMDT 4 plus -TGL-36 "Electronic Flight Bag -CRI F-1/03909" installed r (...) | Master Data List AM001-00-00-PA002 Rev. A dtd. 18. Sep 09-AFM Supplement AM001-00-00-LM001 Rev. A dtd. 30. Sep 09-AMM Supplement AM001-00-00-LM002 Rev. A dtd. 1. Oct 09 or later revisions of the above listed documents approved by EASA | -Application of Electronic Checklist/ECL and WLAN need a separate approval. No external devices are approved to be connected to the display during movement/flight of the aircraft. | 17.06.2013 |
| 10037255 | AERO VODOCHODY AEROSPACE A.S. | | REV. 2 | | A330-201, -202, -203 A330-223, -223F A330-243, -243F A330-301, -302, -303 A330-321, -322, -323 A330-341, -342, -343 | Implementation of Electronic Flight Bag Class 2. This STC installs provisions for Class 2 EFB (navAero T.Bag C ²). | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-25 Amendment 10. The requirements for environmental protection and the associated cert (...) | Master Data List, Ref : AM002-MDL Revision 1, dated 06 October 2011. Instruction for Continued Airworthiness, Ref : AM002-CA, dated 04 October 2011. Aircraft Flight Manual Supplement, Ref : AM002-AM Revision 03, dated 30 September 2011 or later revisions of the above listed documents approved by EASA | None | 17.06.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|----------------------|----------------|----------------------|--|---|---|--|---|---------------|
| 10040961 | AERO VODOCHODY AEROSPACE A.S. | | REV. 2 | | A300 SERIES, A310 SERIES | Implementation of provisions for EFB Class 2. NavAero T.Bag C2 in accordance with Aero Vodochody MDL AM003-MDL Rev 4, Rev 1 of this STC addresses the installation of power switches to the 28V DC bus for safe power off for the EFB-system. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRJ H-01 Special Condition for Enhanced Airworthiness Programme for Aeroplanes system - In (...) | Documents for initial issue of this approval: Aero Vodochody Maintenance Manual Supplement AM003-MM, original issue Aero Vodochody Aircraft Flight Manual Supplement AM003-AM Rev. 2, Aero Vodochody Wiring Diagram Supplement AM003-AM, Rev. 1, Aero Vodochody I (...) | FOR MSN 0554, 0560, 0614 AND 0667, REVISION 1 OF THIS RAPPROVAL IS VALID ONLY UNDER THE FOLLOWING CONDITION: ACCOMPLISHMENT OF INSTALLATION OF SWITCHES ACCORDING TO AERO VODOCHODY MINOR CHANGE, A-STC-Z 0058 FOR MSN0554, A-STC-Z 0059 FOR MSN 0560, A-ST (...) | 17.06.2013 |
| 10043627 | AERO VODOCHODY AEROSPACE A.S. | | REV. 1 | | A340-211, A340-212, A340-213 A340-311, A340-312, A340-313 A340-541, A340-542 A340-642, A340-643 | Implementation of provisions of EFB Class 2 Installation of provisions for detachable Electronic Flight Bag Class 2 on both pilot and co-pilot stations. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Data List AM004-MDL Revision 3, dated 08 Feb 2013; Aircraft Flight Manual Supplement AM004-AM Revision 1, dated 01 Nov 2012; Instructions for Continued Airworthiness AM004-CA Revision 0, dated 20 Aug 2012; Maintenance Manual Supplement AMO (...) | The display, the software and the applications included in the device installed are not covered by this approval. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previ (...) | 17.06.2013 |
| 10046400 | AERO VODOCHODY AEROSPACE A.S. | | REV. 1 | | 747-400 | Implementation of provisions for EFB Class 2 Provisions for use of detachable Class 2 information system on both pilot stations. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 25 at amendment 4, §25.25, §25.27, §25.301(a), §25.303, §25.305(a)(b), (...) | AM005-46-20-EO, Revision -, 2013-04-30 - Engineering Order and Certification Plan (-, <-) > AM005-46-20-MDL, Revision A, 2013-07-11 - Master Data List (-, <-) > AM005-46-20-AM, Revision -, 2013-03-25 - Aircraft Flight Manual Supplement (-, <-) > AM005-46-20-CA, Re (...) | This approval covers only the Class 2 EFB mounting devices and their power and data connection to the aircraft. This approval does not constitute an operational approval for the use of the EFB system. Prior to installation of this design change it must (...) | 30.09.2013 |
| 10029795 | AEROACOUSTICS AIRCRAFT SYSTEMS, INC | | | FAA STC SA02042SE | CESSNA 208 | Aircraft Payload Extender (APE) STOL System for Cessna 208 and 208A in acc. to FAA STC SA02042SE | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certificated noise and/or emissions levels | AASI Master Drawing List No. AA2034, Rev. B, dated December 15, 2009; AASI Installation and Maintenance Manual No. 2014, Rev. B, dated January 6, 2010; AFMS-C208-45, "Aircraft with PT6A-114A (675 SHP) engine", dated January 6, 2010; AFMS-C208-46, "Aircr (...) | Note 1: This modification is also applicable to the following aircraft that have been modified with Wipaire, Inc. STC No. SA149CH (675 SHP PT6A-114A Engine Installation), Model, Serial Number, ..., Note 208, ..., 2080001 through 20800060. Eligible only if mod (...) | 21.04.2010 |
| 10029821 | AEROACOUSTICS AIRCRAFT SYSTEMS, INC | | | FAA STC SA01805SE | 208B | Aircraft Payload Extender (APE) STOL System in acc. to FAA STC SA01805SE | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certificated noise and/or emissions levels | AASI Master Drawing List No. AA1977, Rev. A, dated September 4, 2007; AASI Installation and Maintenance Manual No. 1967, Rev. N/C, dated October 31, 2007; FAA approved Aircraft Flight Manual Supplement AFM-C208-43, dated November 29, 2007 or later revis (...) | The installation must be maintained in accordance with Section 3.0 of Document No. AA1976 | 22.04.2010 |
| 10033826 | AEROACOUSTICS AIRCRAFT SYSTEMS, INC | | | FAA STC SA01239SE | 208 | Fabrication and installation of AeroAcoustics Aircraft Systems (AAS) Aircraft Payload Extender III (APE III) system in acc. with FAA STCSA01239SE. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AASI Master Drawing List No. AA1680, Rev. C, dated January 15, 2010; AASI Installation and Maintenance Manual No. 1679, Rev. D, dated December 20, 2009; AFMS-C208-25, "Landplane with PT6A-114 (600 SHP) engine", Revision C, dated December 9, 2010; AFMS-C208- (...) | This modification is limited to the following landplanes (serial numbers) only: 20800001 through 20800060 (eligible only if modified with Cessna Kit SK-208-12) 20800061 and up; Speed limits: Maneuvering Speed @ B.362 lbs: 143 KIAS; CG Range: Takeoff and Fl (...) | 10.02.2011 |
| 10033827 | AEROACOUSTICS AIRCRAFT SYSTEMS, INC | | | FAA STC SA01180SE | 208 | Fabrication and installation of AeroAcoustics Aircraft Systems (AAS) Aircraft Payload Extender II (APE II) system in acc. with FAA STCSA01180SE. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AASI Master Drawing List No. AA1662, Rev. G, dated January 20, 2010; AASI Installation and Maintenance Manual No. 1663, Rev. F, dated December 20, 2009; AFMS-C208-19, "Landplane with PT6A-114 (600 SHP) engine", Revision B, dated December 9, 2010; AFMS-C208- (...) | Additional limitation: Refer to original FAA STC SA01180SE Serial Number: 20800001 through 20800060 (eligible only if modified with Cessna Kit SK-208-12) 20800061 and up. This modification is also applicable to the following aircraft equipped with Wipline (...) | 10.02.2011 |
| 10016152 | AEROCET, INC. | EASA, IM, A.S. 02354 | | FAA STC SA00137SE | CESSNA 180A CESSNA 180H CESSNA 180J CESSNA 180K | Installation of Aerocet Model 3500 or 3500L seaplane floatation Cessna 180 Aircraft. | | Installation must be in accordance with Master Document List 35-10006 Rev. B dated 21 June 1994 or later Approved Revisions. Operation must be in accordance with Flight Manual Supplement No. A-10018 (180A-180F), A-10019 (180G-180H) and A-10020 (180J, (...) | Those Cessna Model 180G, 180H, 180J and 180K airplanes equipped with the structural modifications defined in Kenmore Air Harbour STCNs, SA649NW, Amendment dated 14 July 1984, may be operated on Aerocet Model 3500 or 3500L floats at gross weight up to (...) | 09.08.2007 |
| 10016153 | AEROCET, INC. | EASA, IM, A.S. 02355 | | FAA STC SA5908NM | CESSNA 185A CESSNA 185B CESSNA 185C CESSNA 185D CESSNA 185E CESSNA A185E CESSNA A185F | Installation of Aerocet Model 3500 or 3500L seaplane floatation Cessna 185 Aircraft. | | Installation must be in accordance with Master Document List 35-10006 Rev. B dated 21 June 1994 or later Approved Revisions. Operation must be in accordance with Flight Manual Supplement No. A-1001 dated 15 March 1995, or later approved revision. | The airplane must incorporate the Cessna approved seaplane kit referred to Section 7 of the Aerocet AFM. The airplane must be equipped with the Cessna rudder centering device/non-linear rudder centering spring system defined on Aerocet Drawing No. 3 (...) | 09.08.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|-------------------|----------------|----------------------|---|---|--|--|---|---------------|
| 10016154 | AEROCET, INC. | EASA.IM.A.S.02356 | | FAA STC SA00003SE | CESSNA P206 CESSNA P206A CESSNA TP206A CESSNA TU206A CESSNA TU206B CESSNA TU206C CESSNA TU206D CESSNA TU206E CESSNA TU206F CESSNA TU206G CESSNA U206A CESSNA U206B CESSNA U206C CESSNA U206D CESSNA U206E CESSNA U206F CESSNA U206G | Installation of Aerocet Model 3500 or 3500L seaplane floats | | FAA approved Aerocet Master Drawing List No. 35-10010, Revision H, dated 26-Jan-2006, or later approved revisionFAA approved Aerocet Aircraft Flight Manual Supplement, Doc. No. A 10011, Revision 1, dated 26-Jan-2006, or later approved revisionFAA ap (...) | | 03.07.2007 |
| 10015658 | AEROCON ENGINEERING COMPANY | EASA.IM.A.S.01635 | | FAA STC ST01825LA | A340-500 | Installation of Cabin Interior Noise Reduction Kit in accordance with Aerocon Engineering Company's Master Document List No. AC25-340-1000 revision C dated August 18, 2005. Validation of FAA STC No. ST01825LA dated August 31, 2005 | | | | 01.02.2006 |
| 10015746 | AEROCON ENGINEERING COMPANY | EASA.IM.A.S.01778 | | FAA STC ST01086LA | JETSREAM 4100 | Installation of Smoke Detection & Fire Suppression Systems for Conversion of Class D Cargo Compartments to Class C | | Aerocon Engineering Master Data List AC26-4100 Revision B or later approved revision.Aerocon Engineering AFM Supplement26-4100-4101 dated December 8 2000 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer | 17.02.2006 |
| 10014554 | AEROCONDOR TRANSPORTES AEREOS SGPS | EASA.A.S.02652 | | | DORNIER DO 228-202K | Installation of Honeywell TCAS system CAS-67A with Diversity Mode S Transponder MST-67A, Control Unit PS-550 and two E-VSI IVA-81A | | EASA Approved Aircraft Flight Manual Supplement No. ATA.021J.081.0502.01, Issue 02, dated November 14, 2006, or later approved revisions | | 21.11.2006 |
| 10014754 | AEROCONDOR TRANSPORTES AEREOS SGPS | EASA.A.S.02940 | | | | Installation of TAWS | | Certification Programme STC ATA0601. Issue: DOAFM - ATA.021J.081.0601.01 | As per AFM | 26.03.2007 |
| 10014780 | AEROCONDOR TRANSPORTES AEREOS SGPS | EASA.A.S.02978 | | | | Installation of Honeywell TAWS Mark VI System with GPS Antenna Honeywell KA-92 and OAT Probe. | | Definition Document ATA.21J.81.0603.002.01, Issue 1.AFM Supplement No. ATA.21J.081.0603.010, Issue 1 | | 08.08.2007 |
| 10014525 | AEROCONSEIL | EASA.A.S.02610 | | | BOEING 757 | Mode S Elementary & Enhanced Surveillance System Installation | | ATS International Master Data List 0249-01-A-2103-R00 datedMay 16 2006 or later approved revision.ATS International AFM Supplement ref 0249-01-A-2301-R00 dated May 16 2006 or later approved revision. | | 01.06.2006 |
| 10014698 | AEROCONSEIL | EASA.A.S.02865 | | | AIRBUS A340 | Installation of Aeromobile (AIRINC/Telenor Joint Venture) GSM Connectivity System. | | 1. Master Data List 0300-01-A-2103-R00;2. Certification Plan 0300-01-A-2101-R00;3. System Technical Specification 0300-01-A-2003-R02;4. Compliance Record 0300-01-A-2104-R01;5. Aircraft Flight Manual Supplement 0300-01-A-2301-R00;6. (...) | 1. The applicability of this STC is limited to the Airbus A340-313 and A340-541 and any other A340 shown to the identical A340-313 MSN 139 or A340-541 MSN 457, through the DOA's process as defined within the Document 0001-01-B-0914 (Classificatio (...) | 05.03.2008 |
| 10014699 | AEROCONSEIL | EASA.A.S.02866 | REV. 3 | | A330 | Installation of Aeromobile GSM Connectivity System onboard Airbus A330AircraftRevision of the EASA STC 10014699 Rev. 2 to remove EWIS limitation | In accordance with the EASA TCDSThe Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval- CRI H-01 "Enhanced Airworthiness Programme for Aeropla (...) | - Master Data List, ref. doc. No. 0313-01-A-2103-R16, Rev. R16, dated16th September 2010- Certification Plan, ref. doc.No. 0313-01-A-2101-F01-R01, Rev. R01, dated 1st February 2010- Compliance Records, ref.doc. No. 0313-01-A-2104-F01-R02, Rev. 02,dat (...) | Not applicable | 17.09.2010 |
| 10014898 | AEROCONSEIL | EASA.A.S.03128 | | | BOEING 777-200 BOEING 777-300 | Installation of Aeromobile GSM connectivity system. | | Master Data List 0333-01-A-2103-R00Certification Plan 0333-01-A-2101-R00Compliance Record 0333-01-A-2104-R00Aircraft Flight Manual Supplement 0333-01-A-2301-R00Aircraft Flight Manual Supplement 0333-01-B-2301-R00Instruction for continued Airworthiness A/W 0333-01-A-2211-F01-R00 | The applicability of this STC is limited to any Boeing 777-200 or B777-300 shown to be identical to Boeing 777-200 MSN27247 or Boeing 777-300 MSN 29063, through the DOA's process as defined within document 0001-01-B-0914 (Classification ofModificati (...) | 26.05.2008 |
| 10015046 | AEROCONSEIL | EASA.A.S.03362 | | | AIRBUS A319-133 | Finish Material Change in Private Area on A319CJ MSN 1468 (This modification concerns the change of fabric on part of the existing panels (including seals) in the private office, washroom and conference room in the A319CJ MSN 1468. | | - Master Data List, ref. No. 7007-03-A-2103-R01, Rev. 01, dated 22.11.07; Certification Plan, ref. No. 7007-03-A-2101-R01, Rev. 01, dated 22.11.07; Compliance Records, ref. No. 7007-03-A-2104-R01, Rev. 01, dated 22.11.07; Instruction for (...) | This STC is limited to aircraft Airbus A319-133 MSN 1468. | 30.11.2007 |
| 10015184 | AEROCONSEIL | EASA.A.S.03564 | REV. 1 | | ATR 42-300/-320/-500 ATR 72-202/-212/-212A | Floor To Floor Nets installation on "Quick Change" ATR42 & 72This STC consists in installing the floor to floor nets instead ofcontainer cargo system. Those nets are installed in the container zonedefined by ATR through their Class E Quickchange con (...) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval-CS 25 Amendment 4The requirements for environmental protection and the associatedcerti (...) | -Master Data List, Ref: 4174-01-A-2103-R02-Weight and Balance Manual Supplement, Ref: 4174-01-A-2305-R01 (forATR42-300 / -320 / -500 aeroplane models)-Weight and Balance Manual Supplement, Ref: 4174-02-A-2305-R01 (forATR72-202 / -212 / -212A aeropla (...) | Limited to those ATR aeroplane models having embodied the following ATRmodifications:For the ATR42-300 / -320Option 1;-ATR Mod 1769 - Modify a/c for Cargo Container Loading, and-ATR Mod 1393 - Reinforce all floor panels to 400 Kg/m2, and-ATR Mod 20 (...) | 22.07.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------|------------------|----------------|----------------------|--|---|---|---|--|---------------|
| 10015238 | AEROCONSEIL | EASA.A.S.03665 | | | AEROSPATIALE CORVETTE | Finish Material Change on Aerospatiale SN 601 #Corvette#This STC concerns the change of finish materials in the cabin such as interior linings, seat foam and cover and floor. | | Master Data List, ref : 4187-01-A-2103-R01 (or any later approved revision of this document, except for referenced document which are approved separately, as ICA for instance) Instruction for Continued Airworthiness, ref : 4187-01-A-2201-R00 | This STC is limited to the SN 601 #Corvette# MSNs : 11, 13,30, 31, and 32.Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification wi (...) | 25.08.2008 |
| 10028688 | AEROCONSEIL | P-EASA.A.S.03382 | | | A318/A319/A320/A321 | Approval of Installation Provisions for the use of EFB Class 2 on AirbusA318/A319/A320/A321 | In accordance with the EASA TCDSThe Certification Basis for the original product remains applicable to this certificate/approvalThe certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | - Master Data List, ref. doc. No. 0316-01-A-2103-R02, Revision R02,dated 3rd February 2010- Airplane Flight Manual Supplement, ref. doc. No. 0316-01-A-2301-R01,Revision R01, dated10th December 2009- Instruction for Continued Airworthiness, ref. doc. (...) | Not Applicable | 03.02.2010 |
| 10031972 | AEROCONSEIL | P-EASA.A.S.03959 | | | A330-200/-300 | Humidification System Installation on A330 aircraft | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval- EASA CS 25 Amendment 4 for areas and parts affected by the change - CRI H-01 Special (...) | - Master Data List doc. Ref. N°0435-01-A-2103-R01, Rev.1 dated 21September 2010.- Instructions for Continued Airworthiness doc. Ref. N°0435-01-A-2211,Rev 00 dated 06 August 2010.or later revisions of the above listed documents approved by EASA | None | 28.09.2010 |
| 10032763 | AEROCONSEIL | | | | 777-300ER | Onboard Mobile Phone system installation on Boeing 777-300ER aircraftprovisioned by Boeing | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalSpecial ConditionCRI H-01 EWISThe certificated noise and/or emissions levels of the or (...) | MDL 0333-02-A-2103-R00 dated 25/11/2010Certification Plan 0333-02-A-2101 rev 01Aircraft Flight Manual Supplement (AFMS)0333-02-A-2301 rev 02Aircraft Maintenance Manual Supplement0333-02-A-2304 initial issueInstruction for Continued Airworthiness (ICA) 0333-02-A-2211 rev. 01 | The applicability of this STC is limited to Boeing 777-31HER aircraft(MSN38983, MSN38984 and MSN35595 | 30.11.2010 |
| 10034165 | AEROCONSEIL | | | | F2000 | FALCON 2000 S/N 226 INTERIOR MATERIAL CHANGESChange consists in refurbishing certain panels- Windows Panels & Valance Panels (lower and upper)- Glass Carbon 250/56%/G874 panels. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalCRI D-119 Flammability testing of aircraft cabin interior usingsurrogate panelsThe requ (...) | Aeroconseil Master Data List doc. N° 4322-01-A-2103-R00 dated March 11th2011or later revisions of the above listed documents approved by EASA | Restricted to S/N 226 | 14.03.2011 |
| 10034950 | AEROCONSEIL | | | | AS 365N3 | Installation of structural provisions (floor fittings) for cabin layoutModification in accordance with AEROCONSEIL Installation ServiceBulletin 2018-01-A-3101. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | -Aeroconseil Master Data List (MDL) 2018-01-A-2103-R00-Aeroconseil Installation Service Bulletin 2018-01-A-3101-R00-Aeroconseil Instruction for Continued Airworthiness (ICA)2018-01-A-2211-R00 | None | 16.05.2011 |
| 10035182 | AEROCONSEIL | | | | A300 B4-601, A300 B4-603, A300 B4-620, A300 B4-622, A300 B4-622 R, A300 B4-605 R, A300 C4-605 R VARIANT F, A300 C4-620, A300 F4-605 R, A300 F4-622 R | Aeroconseil STC 0490-1 Aircell ST3100 SATCOM Installation on A300-600s.This STC covers the installation of a Satellite Telephone SystemTransceiver, a Telephone adapter with Display, a Base cradle for theCordless Handset and a Satellite antenna in ac (...) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval- EASA - Certification Specification CS 25 Amdt 4- Special condition S/C H01. (Enhanc (...) | -A300-600 Flight Manual Supplement Ref.0490-01-A-2301 Rev.00- Instruction for Continued Airworthiness Document Ref.0490-01-A-2211 - Compliance Records Ref.0490-01-A-2104 Rev.01- Master Data List Ref.0490-01-A-2103 Rev.01 or later revisions of the above listed documents approved by EASA. | None. | 01.06.2011 |
| 10035880 | AEROCONSEIL | | | | B777-200, B777-200LR B777-300, B777-300ER | 0555-01: Provisions of OnAir GSM/Internet System on B777 | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.-CS 25 Amdt. 4- CRI H-1 - ICA on EWISThe certificated noise and/ or emissionslevels of (...) | 0555-01 "Provisions of OnAir GSM/Internet System on B777"-Master Document List 555-01-A-2103-R01 dtd. 04.Jul.2011-Instructions for Continued Airworthiness 555-01-A-2211-R00 dtd.04.Jul.2011 for later revisions of the above listed documents approved by EASA | None | 25.07.2011 |
| 10036272 | AEROCONSEIL | | REV. 1 | | A340-311/-312/-313 | A340 Cabin Reconfiguration & Zodiac Sit In-Flight EntertainmentInstallationThe modification consists in reconfiguring the A340 cabin compartment byreplacing the economy class seats, from 218 Y/C seats to 215 Y/C seatswith Sigma 3537 seriesseats, and (...) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS 25 Amdt 4Except for the following requirements, which are as per TCDS EASA.A.015:-2 (...) | Master Data List, ref: 7048-01-A-2103-R02Instruction for Continued Airworthiness, ref: 7048-01-A-2211-R00or later revisions of the above listed documents approved by EASA | 1. Aeroconseil Modification, ref: MOD 7049-01, must be installed(->,-<)-being a pre-requisite to this STC.2. This STC islimited to the A340-300 MSNs 197 and 378 | 09.08.2012 |
| 10037524 | AEROCONSEIL | | | | A330-301/-302/-303 A330-321/-322/-323 A330-341/-342/-343 | Panasonic eXPhone System activation on A330-300 with eX2 IFEThis STC installs a Panasonic eXPhone System allowing, in conjunctionwith other devices, the use of cellphones during flight. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.-CS 25 Amendment4-CRI H-01 Issue 1 - Enhanced Airworthiness Programme for AeroplaneSys (...) | Master Data List, Ref : 0496-02-A-2103-R00, dated 01 December 2011Instruction for Continued Airworthiness, Ref : 0496-02-A-2211-R01, dated30 November 2011Aircraft Flight Manual Supplement, Ref : 0496-02-A-2301-R00, dated 05October 2011Instructions f (...) | The following modifications are prerequisites to the installation ofthis STC:-Airbus Minor Change ref: 200190 - Approved under Airbus DOAEASA.21.J.031, on 22 February 2011.-Delta Engineering STC ref: ODA-2234-01 - EASA Approval 10036548 Rev 1or later (...) | 02.12.2011 |
| 10037893 | AEROCONSEIL | | | | 747-100, -200, -200F | Installation of a CAS-81 TCAS II system and Dual TRA-87 Mode STransponders in accordance with Aeroconseil Master Drawing List0520-01-A-2103-R00 or later approved revisions. | The Certification Basis for the original product remains applicable tothis certificate/ approvalThe requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval | Aeroconseil AFM supplement no 0520-01-A-2301-R00or later revisions of the above listed documents approved by EASA | None | 04.01.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------|--------------|----------------|----------------------|--|---|--|---|---|---------------|
| 10038574 | AEROCONSEIL | | | | A330-200, -300 | A330 Tape Video Reproducer Unit Installation The change consist in replacement of each MAS Video Tape Player (VTP) by new digital videoreproducer units manufactured by Goodrich installed in the aircraft existing Video Centro Center. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: A330 EASA TCDS A.004 The requirements for environmental protection and the associated cder (...) | Certification Plan 4311-01-A-2101-R02 dated 24th February 2012. Master Data List 4311-01-A-2103-R01 dated 27th February 2012. Instructions for Continued Airworthiness 4311-01-A-2211-R00 dated 17th February 2012. Aircraft Configuration Index Document (AC (...)) | this change is limited to A330-200 and A330-300, providing that the VCCP/N RDA A902251 is installed, and still in the configuration approved by the aircraft manufacturer (no other STCs modifying the VCC before this change). | 27.02.2012 |
| 10038663 | AEROCONSEIL | | REV. 1 | | 777-200, 777-200LR 777-300, 777-300ER | 0411-01: Installation/activation of OnAir GSM/Internet System on B777 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-25 Amdt. 4-CRI F-01 GSM System-CRI F-02 Hardware (ref AEC STC 0333-01)-CRI F-03 Sof (...) | Master Data List 0411-01-2103 R00 dtd. 6.March 2012-AFM Supplement 0411-01-A-2301-R01 dtd. 17. Jan 2012-Instructions for Continued Airworthiness 0411-01-A-2211-R00 dtd.29.Feb 2012 or later revisions of the above listed documents approved by EASA | Per AFM Supplement 0411-01-A-This modification is limited as stated below except similarity demonstrated per Document 0001-01-B-0914 (Classification of Modification/Change Procedure for Connectivity System).STC effectivity: Boeing 777-300ER MSN 33377, 3 (...) | 08.03.2012 |
| 10038678 | AEROCONSEIL | | | | B777-200 | 0527-01: eXPhone System Installation on B777-200 aircraft. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:- CS 25 Amdt 4-CRI F01 GSM System (ref AEC STC 0333-01)- CRI F02 Hardware (ref AEC ST (...)) | MDL 0527-01-2103-R00 dtd. 6.Mar 2012-AFM Supplement 0527-01-A-2301-R00 dtd. 17.Feb 2012-Instructions for Continued Airworthiness 0527-01-A-2211-R00 dtd. 10.Nov.2011 or later revisions of the above listed documents approved by EASA. | Per AFM Supplement 0527-01-A-2301.This modification is limited to Boeing 777-200 MSN29953, except similarity demonstrated per Document 0001-01-B-0914 (Classification of Modification/Change Procedure for GSM connectivity System). | 07.03.2012 |
| 10038812 | AEROCONSEIL | | REV. 1 | | A330-301 / -302 / -303 A330-321 / -322 / -323 A330-341 / -342 / -343 | Panasonic eXPhone System activation on A330-300 with eXconnect. This STC installs a Panasonic eXPhone System allowing, in conjunction with other devices, the use of cellphones during flight. This revision 01 extends the STC applicability to a second aircraft configuration. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:- CS 25 Amendment 4- CRI H-01 Issue 1 - Enhanced Airworthiness Programme for AeroplaneS (...) | Master Data List, Ref : 0496-01-A-2103-R01, dated 23 July 2012; Instruction for Continued Airworthiness, Ref : 0496-01-A-2111-R03, dated 06 July 2012; Aircraft Flight Manual Supplement, Ref : 0496-01-A-2301-R03, dated 05 June 2012; Weight and Balance Report (...) | none | 21.08.2012 |
| 10038876 | AEROCONSEIL | | | | B777-200 | 0542-01: WiFi internet access system installation on B777-200 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:- CRI F-01 / 10010868 - WiFi installation-CRI H-01: ICA on EWIS (ref AEC STC 0555-01) | MDL 0542-01-2103-R00 dtd. 15.Mar 2012-AFM Supplement 0542-01-A-2301-R00 dtd. 15.Mar 2012; Instructions for Continued Airworthiness 0542-01-A-2211-R00 dtd. 15.Mar 2012; or later revisions of the above listed documents approved by EASA. | - per AFMS 0542-01-A-2301- Airworthiness Limitations per ICA Document 0542-01-A-2211 amendmentary. This modification is limited to Boeing 777-200 MSN29953, except similarity demonstrated per Document 0001-01-B-0914 (Classification of Modification Change Procedure for GSM connectivity System). | 21.03.2012 |
| 10039705 | AEROCONSEIL | | | | A330-201 / -202 / -203 A330-223 / -243 | Panasonic eXPhone System activation on A330 with eXConnect and Rack P/NRD-AA903023 This STC installs a Panasonic eXPhone System allowing, in conjunction with other devices, the use of cellphones during flight. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:- CS 25 Amendment 4, - CRI H-01 Issue 1 - Enhanced Airworthiness Programme for Aeropla (...) | Master Data List, Ref : 0585-01-A-2103-R00, dated 15 May 2012; Instruction for Continued Airworthiness, Ref : 0585-01-A-2211-R00, dated 06 April 2012; Aircraft Flight Manual Supplement, Ref : 0585-01-A-2301-R00, dated 16 May 2012; or later revisions of the above listed documents approved by EASA. | None. | 18.05.2012 |
| 10040592 | AEROCONSEIL | | | | A330-201 / -202 / -203 A330-223 / -243 | Panasonic eXPhone System Activation on A330 with eX2 IFE and eXconnect. This STC installs a Panasonic eXPhone System allowing, in conjunction with other devices, the use of cellphones during flight. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:- CS 25 Amendment 4, - CRI H-01 Issue 1 - Enhanced Airworthiness Programme for Aeropla (...) | Master Data List, Ref : 0496-03-A-2103-R00, dated 11 July 2012; Instruction for Continued Airworthiness, Ref : 0496-03-A-2211-R00, dated 10 July 2012; Aircraft Flight Manual Supplement, Ref : 0496-03-A-2301-R00, dated 10 July 2012; Weight and Balance Rep (...) | None. | 12.07.2012 |
| 10041655 | AEROCONSEIL | | | | 777-200LR & 777-300ER | 0465-01: eXPhone System Activation on B777-200LR/300ER | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 25 Amdt 4-CRI-F01: PANASONIC EXPHONE SYSTEM INSTALLATION-CRI-F02: HARDWARE-CRI-F03: SO (...) | MDL 0465-01-A-2103-R01 dtd. 3.Oct 12AFMS 0465-01-A-2301-R00dtd. 23.Jul 12ICA 0465-01-B-2211-R00 dtd. 26.Jul 12or later revisions of the above listed documents approved by EASA | per AFMS 0465-01-A-2301 | 04.10.2012 |
| 10042771 | AEROCONSEIL | | | | A320-111,-211,-212,-214,-215 A320-216,-231,-232,-233 | Installation of SICMA Dragonfly seats for Indigo on A320 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 25 Amdt 4 (except for §25.785 applicable at JAR 25 Amdt 11) The requirements for env (...) | ..Certification Plan n.4490-01-A-2101-R00 dated 05/09/2012-..Compliance Records n.4490-01-A-2104-R01 dated 07/12/2012-..Master Data List n.4490-01-A-2103-R00 dated 14/12/2012-..Instructions for Continued Airworthiness n.4490-01-A-2211-R00 dated 11/ (...) | none | 18.12.2012 |
| 10043930 | AEROCONSEIL | | | | AS 365 N3 | White Strobe Light Installation on the Underside Fuselage | The Certification Basis for the original product remains applicable to this certificate/ approval the following paragraph(s) at a later amendment: CS 29 Amdt.2 paragraphs:29.1, 29.27, 29.29, 29.601, 29.603, 29.605, 29.607, 29.609, 29.610,29.777, 29.13 (...) | Rotorcraft Flight Manual Supplement, document no. 2027-01-A-2301-R02 dated 26 April 2012 or later revisions approved by EASA. Master Certification Data Sheet, document no. 2027-01-A-2115-R03 dated 04 March 2013 Instructions for Continued Airworthiness, (...) | - Modification limited to AS365N3 with Serial Numbers SN6687 and SN6690. The use of the white strobe light system with Night Vision Image Systems (NVIS) or Night Vision Goggles (NVG) is forbidden. - The operation of the white strobe light on the under (...) | 06.03.2013 |
| 10043931 | AEROCONSEIL | | | | AS 365 N3 | Passenger Address System Installation. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:- the following paragraph(s) at a later amendment: CS 29 Amdt. 2 paragraphs: 29.1, 29.27, 29.29, 29.607, 29.610, 29.777, 29.831, 29. (...) | Master Certification Data Sheet, document no. 2026-01-A-2115-R03, Revision 03; Instructions for Continued Airworthiness, document no. 2026-01-A-2211-R02, Revision 02; Service Bulletin, document no. 2026-01-A-3101-R02, Revision 02; or later approved revisions of the above listed documents. | Modification limited to AS365N3 with Serial Numbers SN6687 and SN6690. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repa (...) | 06.03.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|----------------|----------------|----------------------|---|---|---|--|---|---------------|
| 10044536 | AEROCONSEIL | | | | A318-111 / -112 / -121 / -122 A319-111 / -112 / -113 / -114 A319-115 / -131 / -132 / -133 A320-111 / -211 / -212 A320-214 / -215 / -216 A320-231 / -232 / -233 A321-111 / -112 / -131 / -211 A321-212 / -213 / -231 / -232 | Mechanical and electrical provision installation for Personal Tablet/Computer. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25Amdt. 4 for the paragraphs affected by the change. The requirements for environmental protection and the associated de (...) | Master Data List 0621-01-A-2103-R00 dated 16/04/2013 Certification Plan 0621-01-A-2101-R02 dated 12/04/2013 Compliance Record List 0621-01-A-2014-R01 dated 11/04/2013 Instructions for Continued Airworthiness 0621-01-A-2211-R01 dated 11/04/2013 or later r (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 18.04.2013 |
| 10044938 | AEROCONSEIL | | | | AS 365 N3 | Installation of ADELTA - CPI 503-16 Crash Position Indicator System | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 29 Amdt. 2 paragraphs: 29.1, 29.27, 29.29, 29.251, 29.603, 29.605, 29.6 (...) | Master Data List, document no. 2000-01-A-2103-R02, dated 25 April 2013 Instructions for Continued Airworthiness, document no. 2000-01-A-2211-R01 dated 06 April 2012 | Modification limited to AS365N3 with Serial Numbers SN6687 and SN6690. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or re (...) | 21.05.2013 |
| 10045396 | AEROCONSEIL | | | | A319-111/-112/-113/-114/-115 A319-131/-132/-133 A320-111/-211/-212/-214/-215 A320-216/-231/-232/-233 A321-111/-112/-131/-211/-212 A321-213/-231/-232 | Installation of ACS's TCAS 2000 (version 7.1) software upgrade | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification Plan 0594-01-A-2101-R00 dated 27/03/2013 Master Data List 0594-01-A-2103-R00 dated 11/06/2013 Compliance Records 0594-01-A-2104-R00 dated 27/03/2013 Instruction for Continued Airworthiness 0594-01-A-2211-R01 dated 25/02/2013 or later revisi (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 20.06.2013 |
| 10045443 | AEROCONSEIL | | | | 777-300ER | eXPhone System activation on B777-300ER with VVIP cabin interior [Installation of eXPhone dedicated equipment and activation of their-flight GSM connectivity system] | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List, Aeroconseil document No. 0616-01-A-2103, Revision 0, dated 14 June 2013. Aircraft Flight Manual Supplement, Aeroconseil document No. 0616-01-A-2301 (>,<,>) Revision 1, dated 11 June 2013 Instruction for Continued Airworthiness, Ae (...) | Per Instruction for Continued Airworthiness, Aeroconseil document No. 0616-01-A-2211 this modification is limited to the Boeing B777-200 MSN36563, with VVIP cabin interior, except similarity demonstrated per Aeroconseil document 0001-01-B-0914 (Classif (...) | 25.06.2013 |
| 10045473 | AEROCONSEIL | | | | A330-201/-202/-203/-223/-243 A330-223F/-243F/-301/-302/-303 A330-321/-322/-323 A330-341/-342/-343 | Mechanical & Electrical provision for personal tablet computer for A330 aircraft. This STC deals with the installation of provision both mechanical (mounting assembly support on lower frame of each cockpit sliding window) and electrical (DC/USB converter) for Personal Tablet/Computer. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification plan: 0632-01-A-2101-R02 Master Data List: 0623-01-A-2103-R00 ICA: 0632-01-A-2211-R02 AFM Supplement: 0632-01-A-2301-R00 or later revisions of the above listed documents approved by EASA | The window frame must be inspected at 34230 FC and 16475 FC inspection interval (report 632-01-A-3403-F02). Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously in (...) | 27.06.2013 |
| 10038123 | AEROCONTROLEX GROUP, INC. | | | FAA STC ST02798CH | 1. B767-200/-300/-400ER 2. B777-200/-300/-300ER/-200LR | Installation of an AeroControlex Potable Water Level Sensor Assembly P/N8430-000000 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-1 ICA on EWIS | 8430-000000 DL Rev. Original, no date 80-0215 Installation Instructions Rev. C, dtd. 27-Jan-2010 038-10-08 Instructions for Continued Airworthiness Rev. 2, dtd. 18-Jun-09 938-10-09 Instructions for Continued Electrical Airworthiness Rev. 2 dtd. 18-Jun-09 (...) | None | 20.01.2012 |
| 10045881 | AERODATA AG | | | | B300 | Installation of Aerodata Flight Inspection System AD-AFIS-261. | The Certification Basis for the original product remains applicable for items not affected by the modification, for items affected by the modification the following new Certification Basis is elected to comply with: CS-23 Amdt. 3, except 23.561 and 23. (...) | Installation in accordance with Master Document List, Aerodata Doc. No. MDL-A/C-840, Rev. 1, dated 19 July 2013. Aircraft Flight Manual Supplement, Aerodata Doc. No. AFM-134, Rev. 0, dated 05 June 2013. Aircraft Maintenance Manual Supplement, Aerodata Do (...) | Only applicable for Serial FL-721 | 26.07.2013 |
| 10046756 | AERODATA AG | | | | B300 | Installation of Aerodata "AD-MS-1000" Mission System. The installation consists of the installation of a non-essential sensor system underneath the belly with 360° Radar into Beechcraft certified and installed radom, EO/IR turret into Beechcraft certif (...) | The Certification Basis for the original product remains applicable for items not affected by the modification, for items affected by the modification the following new Certification Basis is elected to comply with: CS-23 Amdt. 3, except 23.561 and 23. (...) | Installation in accordance with Master Document List, Aerodata Doc. No. MDL-A/C-836, Rev. 1, dated 10 September 2013. Aircraft Flight Manual Supplement, Aerodata Doc. No. AFM-135, Rev. 2, dated 11 September 2013. Aircraft Maintenance Manual Supplement, A (...) | Only applicable for Serial FL-795. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon (...) | 18.10.2013 |
| 10014029 | AERO-DIENST GMBH & CO. KG | EASA.A.S.01095 | | | LEARJET 60 | Installation of a Cabin Entertainment System with Airshow 4X0. | Aerodienst engineering Order N0 203-LR, Rev. IR, or later EASA-approved revisions. | For certification basis refer to FAA TCDS A10CE plus FAR 25.1309 at Amdt. 25-41 and FAR 25.1529 at Amdt. 25-54 and plus JAA TGL No.17, IFE Systems. | 28.04.2005 | |
| 10014101 | AERO-DIENST GMBH & CO. KG | EASA.A.S.01203 | | | LEARJET 60 | Installation of an Iridium Satellite ICS-200 System. | Aerodienst Engineering Order No 202-LR, Rev. IR, or later EASA-approved revisions. | For certification basis refer to FAA TCDS A10CE plus FAR 25.1309 at Amdt. 25-41 and FAR 25.1529 at Amdt. 25-54. | 28.04.2005 | |
| 10014132 | AERO-DIENST GMBH & CO. KG | EASA.A.S.01242 | | | HAWKER HS 125-700B | Installation of Traffic Alert Collision and Avoidance System Collins TCAS 4000 | Engineering Order Doc No 171-HW Rev 1 or later EASA approved revision Airplane Flight Manual Supplement Doc No AG352-071 Rev IR or later EASA approved revisions | | 30.05.2005 | |
| 10014221 | AERO-DIENST GMBH & CO. KG | EASA.A.S.01383 | | | BAE 125 SERIES 800A BAE 125 SERIES 800B | Installation of a L-3 F1000 SSFDR | Engineering Order Doc. No. 210-HW Rev. IR or later approved Revisions Airplane Flight Manual Supplement L-3 F1000 SSFDR, Doc. No. AG542-071 Rev IR or later EASA approved Revisions Instruction for Continuous Airworthiness Doc. AG542-085 | | 10.11.2005 | |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|----------------|----------------|----------------------|--|---|---------------------|---|--|---------------|
| 10014222 | AERO-DIENST GMBH & CO. KG | EASA.A.S.01384 | | | BAE 125 SERIES 800A BAE 125 SERIES 800B | Installation of a Rockwell Collins TCAS 4000 with Dual TDR-94D and Dual RTU-4210 System | | Engineering Order Doc. No 206-HW Rev. IR or later EASA approved RevisionsAirplane Flight Manual Supplement/Collins TCAS4000 with Dual TDR-94D and Dual RTU-4210. Doc. No. AG530-071 Rev IR or later EASA approved RevisionsInstruction for Continuous Airworthiness Doc. AG530-085 | | 10.11.2005 |
| 10014223 | AERO-DIENST GMBH & CO. KG | EASA.A.S.01385 | | | BAE 125 SERIES 800A BAE 125 SERIES 800B | Installation of an Universal Avionics System Corporation Flight Management System UNS-1L | | Engineering Order Doc. No. 208-HW Rev. IR or later EASA approved RevisionsAirplane Flight Manual Supplement/UASC UNS-1LFMS. Doc. No. AG573-071 Rev IR or later EASA approved RevisionsInstruction for Continuous Airworthiness Doc. AG537-85 | | 10.11.2005 |
| 10014224 | AERO-DIENST GMBH & CO. KG | EASA.A.S.01386 | | | BAE 125 SERIES 800A BAE 125 SERIES 800B | Installation of an UASC CVR-30B | | Engineering Order Doc. No. 209-HW Rev. IR or later EASA approved RevisionsAirplane Flight Manual Supplement/UASC CVR-30B, Doc. No. AG540-071 Rev IR or later EASA approved RevisionsInstruction for Continuous Airworthiness Doc. AG540-085 | | 10.11.2005 |
| 10014382 | AERO-DIENST GMBH & CO. KG | EASA.A.S.01666 | | | LEARJET 25C | Collins TCAS-4000 with Dual TDR-94D System Installation | | Engineering Order No.: 212-LR Rev. IR with Appendix A Airplane Flight Manual Supplement AG524-071, Revision IR | Certification Basis of Change: FAR 25 as defined in TCDS A10CE issue 55, additionally JAA Leaflet No. 8, Rev. 2 (ACAS II), JAA Leaflet No. 13, Rev. 1 (Modes S ELS) | 03.03.2006 |
| 10014385 | AERO-DIENST GMBH & CO. KG | EASA.A.S.01671 | | | HAWKER HS 125-700B | Installation of Honeywell EGPWS Mark VIII System and interface to Honeywell KMD-850 MFD via a KAC-502 EGPWS Module. | | Installation in accordance with Aero-Dienst Engineering Order ref 205-HW, Rev. IR, dated 06. Feb. 2006, or later approved revision. To be operated in accordance with Aero-Dienst Airplane Flight Manual Supplement ref AG516-071, Rev IR, dated 10 April 2006, or later approved revision. | None. | 19.04.2006 |
| 10014386 | AERO-DIENST GMBH & CO. KG | EASA.A.S.01672 | | | HAWKER HS 125-700B | Installation of Honeywell ART 2000 Weather Radar System and Honeywell KMD-850 MFD. | | Installation in accordance with Aero-Dienst Engineering Order ref. 215-HW, Rev IR, dated 06. Feb. 2006, or later approved revision. To be operated in accordance with Aero-Dienst Airplane Flight Manual Supplement ref AG566-071, Rev IR, dated 10 April 2006, or later approved revision. | None. | 19.04.2006 |
| 10014547 | AERO-DIENST GMBH & CO. KG | EASA.A.S.02637 | | | LEARJET 55 | Cockpit Voice Recorder CVR-30B Installation (Universal Avionics System Corp.) | | Engineering Order, 220-LR, Rev. IR, or later approved revisionAFM Supplement, AG581-071, Rev. IR, or later approved revision | | 24.07.2006 |
| 10014548 | AERO-DIENST GMBH & CO. KG | EASA.A.S.02638 | | | LEARJET 55 | Solide State Flight Data Recorder F1000 Installation (L3-AR) | | Engineering Order, EO 219-LR, Rev. IR, or later approved revisionAFM Supplement AG583-071, Rev. IR, or later approved revision | | 24.07.2006 |
| 10014560 | AERO-DIENST GMBH & CO. KG | EASA.A.S.02659 | | | LEARJET 55 | FMS UNS-1L Installation (Universal Avionics System Corp.) | | Engineering Order, 218-LR, Rev. IR, or later approved revisionAFM Supplement, AG586-071, Rev. A, or later approved revision | | 24.07.2006 |
| 10014634 | AERO-DIENST GMBH & CO. KG | EASA.A.S.02777 | | | LEARJET L-55 | Installation of TAWS / RMI ST3400 (Sandel) System | | Certification Program E/O 221-LR, rev. 2AFM Supplement AG601-071, rev. A | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 23.01.2007 |
| 10014662 | AERO-DIENST GMBH & CO. KG | EASA.A.S.02819 | | | LEARJET 31A | Enhanced Surveillance Mode S Transponder Upgrade (Honeywell) | | EO 222-LR, rev. 1Certification Compliance Sheet AG599-075, rev. AAFM Supplement AG599-071, rev. IRInstructions for Continued Airworthiness AG599-085, rev. A | | 28.03.2007 |
| 10014669 | AERO-DIENST GMBH & CO. KG | EASA.A.S.02828 | REV. 1 | | HAWKER HS 125-700B | FMS UNS-1L System Installation (UASC) | | Installation in accordance with Aero-Dienst Engineering Order ref 223-HW, Rev. 1, dated 21. June 2007, or later approved revision. To be operated in accordance with Aero-Dienst Flight Manual Supplement ref AG623-071, Rev A or later approvedrevision. | | 07.01.2008 |
| 10014700 | AERO-DIENST GMBH & CO. KG | EASA.A.S.02869 | | | LEARJET 31 | TCAS-4000 / Dual TDR-94D Mode S Transponder Systems Installation (Collins) | | EO 225-LR, rev. 2Certification Compliance Sheet AG626-075, rev. AAFM Supplement AG626-071, rev. IRInstructions for Continued Airworthiness AG626-085, rev. A | | 26.03.2007 |
| 10014706 | AERO-DIENST GMBH & CO. KG | EASA.A.S.02878 | | | LEARJET 31 | FMS UNS-1L System Installation (UASC) | | Engineering Order No. 226-LR, Rev. 1, dated 24.01.2007 or later approved revisions. Aircraft Flight Manual Supplement AG627-071, Rev. IR or later approved revisions. | Certification basis of the change FAR 25, additionally JAA Leaflet No. 10, Precision (RNAV), AMC 20-4 (Basic RNAV), AMC20-5 (GPS). | 21.03.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|----------------|----------------|----------------------|------------------------------|--|---|--|---|---------------|
| 10014755 | AERO-DIENST GMBH & CO. KG | EASA.A.S.02942 | | | BAE 125 SERIES 800B | TAWS ST3400 System Installation (Sandel) | | Installation in accordance with Aero-Dienst Engineering Order ref 229-HW, Rev. IR dated 17. Jan. 2007, or later approved revision. To be operated in accordance with Aero-Dienst Airplane Flight Manual Supplement ref AG636-071, Rev IR or later approved revision. | | 13.04.2007 |
| 10014955 | AERO-DIENST GMBH & CO. KG | EASA.A.S.03223 | | | LEARJET 35A | MFD-640 with WX Controller RP-1008 Installation (UASC) | | AFM Supplement AG690-071, rev. IR (or later approved revisions) | | 04.12.2007 |
| 10014956 | AERO-DIENST GMBH & CO. KG | EASA.A.S.03224 | | | LEARJET 35A | Cockpit Voice Recorder CVR-120 Installation (UASC) | | AFM Supplement AG691-071, rev. A (or later approved revisions) | | 04.12.2007 |
| 10014962 | AERO-DIENST GMBH & CO. KG | EASA.A.S.03233 | | | LEARJET 35A | Dual VHF NAV VIR-30/DME-40 replacement by VIR-32/DME-42 with Dual CTL-32 installation (Collins) | | AFM Supplement AG692-071, rev. A (or later approved revisions) | N/A | 04.12.2007 |
| 10014968 | AERO-DIENST GMBH & CO. KG | EASA.A.S.03247 | | | LEARJET 35A | Enhanced Surveillance Mode S Transponder modification (Collins) | | AFM Supplement AG693-071, rev. A (or later approved revisions) | | 04.12.2007 |
| 10014969 | AERO-DIENST GMBH & CO. KG | EASA.A.S.03248 | | | LEARJET 35A | Centre Pedestal Enlargement / Nose Compartment Rearrangement. | | N/A | | 04.12.2007 |
| 10015051 | AERO-DIENST GMBH & CO. KG | EASA.A.S.03370 | | | LEARJET 60 | GPWS to EGPWS System Upgrade (Honeywell) | | AFM Supplement AG705-071, rev. IR (or later approved revisions) | | 21.02.2008 |
| 10015150 | AERO-DIENST GMBH & CO. KG | EASA.A.S.03511 | | | LEARJET 35A | Flight Crew Intercom System (NAT) | | As per Certification Program 243-LR, Rev. IR (or later approved revisions), AFM Supplement AG711-071 Rev IR (or later approved revision). | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 10.07.2008 |
| 10015233 | AERO-DIENST GMBH & CO. KG | EASA.A.S.03658 | | | LEARJET 35A | Solid State Flight Data Recorder SSFDR Installation (L-3) | | As detailed in Certification Program AG748-080, rev. IR (or later approved revisions) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 04.11.2008 |
| 10015234 | AERO-DIENST GMBH & CO. KG | EASA.A.S.03659 | | | LEARJET 35A | Cockpit Voice Recorder CVR-30B Installation (UASC) | | As detailed in Certification Program AG751-080, rev. IR (or later approved revisions) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 04.11.2008 |
| 10015272 | AERO-DIENST GMBH & CO. KG | EASA.A.S.03754 | | | LEARJET 55 | FMS Upgrade to PRNAV Compliance (UASC) | | As detailed in Certification Program AG794-080, rev. IR (or later approved revisions) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 02.02.2009 |
| 10036586 | AERO-DIENST GMBH & CO. KG | | | | LEARJET 35A | Installation of Honeywell (Allied Signal) Enhanced Ground Proximity Warning MK VIII System | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification Programme 105-LR Rev IR dated June 21, 2011 and Engineering Order No.: 105-LR.056.01 Rev 6 dated June 21, 2011 or later revisions of the above listed documents approved by EASA | None | 21.09.2011 |
| 10043017 | AERO-DIENST GMBH & CO. KG | EASA.A.S.01204 | | | LEARJET 60 | Installation of a SSFDR F1000 System. | | Aerodienst Engineering Order No 197-LR, Rev. IR, or later EASA-approved revisions. | 1. The certification basis refers to FAA TCDS A10CE plus FAR 25.1309 at Amdt. 25-41 and FAR 25.1529 at Amdt. 25-542. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Descripti (...) | 28.04.2005 |
| 10043082 | AERO-DIENST GMBH & CO. KG | EASA.A.S.03831 | | | 35A | Dual SN4500 EHSI Installation (Sandel) | | As detailed in Engineering Order EO 250-LR, rev. 1 (or later approved revisions) | none | 26.05.2009 |
| 10043084 | AERO-DIENST GMBH & CO. KG | EASA.A.S.03801 | | | LEARJET 35A | Dual FMS UNS-1Lw System Installation (UASC) | | As detailed in Engineering Order EO 249-LR, rev. 1 (or later approved revisions) | none | 26.05.2009 |
| 10043246 | AERO-DIENST GMBH & CO. KG | | REV. 1 | | BD 100-1A10 (CHALLENGER 300) | Installation of a Thrane and Thrane Aviator 300 Swift BroadBand -satcom system with WLAN capability. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Engineering Order 264-CL.056.01 rev 1 dated 28/12/2012. Instructions for Continued Airworthiness ref. AG959-085 rev IR, Airplane Flight Manual Supplement AFMS AG959-071 rev IR or later revisions of the above listed documents approved by EASA. | Refer to AFMS AG959-071 rev IR. | 16.01.2013 |
| 10043837 | AERO-DIENST GMBH & CO. KG | | REV. 1 | | B300 | Alternate Position for ACAS and XPDR Antennas. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Engineering Order: 265-BE.056.01 Rev. 2, or later approved revisions. Instructions for Continued Airworthiness: AG962-085 IR, or later approved revisions. Declaration of Compliance for Project 265-BE 08.02.2013. | Restricted to S/N FL-101, FL-130 - LBA STC SA 1356 must have been already installed. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change (...) | 21.02.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------|------------------|----------------|----------------------|--|---|---|---|---|---------------|
| 10014006 | AERODISA S.L. | EASA.A.S.01043 | | | EMBRAER EMB-120 | Installation TCAS II Honeywell TCAS II TPU 67A Processor and Configuration Module with lower and upper ANT 67A Antennas, dual IVA-81D TA/RA/VI System Displays, one MST67A Mode S Transponder (OPTION 1) or two MST67A Mode S Transponder (OPTION 2) with lower and upper KA60 Omnidirectional Antennas and | | - Doc No. AED2004-01M_MDL "Master Data List", Ed 0, dated 20 April 2006- Doc No. AED2004-01M_PCA "Plan de Comprobación de Aeronavegabilidad", Ed 0 Installation instructions in accordance with- Doc No. AED2004_01M_D01 "Installation Manual" Edición 00 - Honeywell Installation Manual Edición 00- (...) | Subject to the conditions, limitations and procedures contained in Doc No. AED2004_01M_D03_OP1 "Flight Manual Supplement for 1 transponder installation", Rev 0 or AED2004_01M_D03_OP2 "Flight Manual Supplement for 2 transponders installation", Rev. 0 | 31.07.2006 |
| 10014051 | AERODISA S.L. | EASA.A.S.01133 | | | FAIRCHILD SWEARINGEN SA 227 SE FAIRCHILD SWEARINGEN SA 227 BC | TCAS II installation on Fairchild SA 227Honeywell TCAS II TPU 67A Processor and Configuration Module with lower and upper ANT 67A Antennas, dual IVA-81D TA/RA/VI System Displays, two MST67A Mode S Transponders with lower and upper KA60 Omnidirectional (...) | | - Doc N° AED2004-02M_PCA "Plan de Comprobación de Aeronavegabilidad", Ed 0 Installation Instructions in accordance with - Doc N° AED2004-02M_D01 "Installation Manual" Edición 00 - Honeywell Installation Manual Manual 006-05340-0006 CAS67A TCAS I (...) | Subjected to the conditions, limitations and procedures contained in Doc N° AED2004_02M_D03 "Flight Manual Supplement", Rev 0 | 04.09.2006 |
| 10014286 | AERODISA S.L. | EASA.A.S.01523 | | | DC-8-62 SERIES | Revision to the cargo modification on DC-8-62. Compliance with FAA AD 2001-26-04 & AD 2001-26-01. | | Documentation referred in Doc. AED2005-01M Issue 01, April 2006.Maintenance Manual Supplement AED2005-01M_SMM Issue 01, April 2006. | Limited to DC-8-62 series airplanes which have been converted from a passenger to a cargo-carrying ("freighter") configuration by FAA STC No. SA1802SO, but with a deviation in the main deck cargo 9g crash barrier that was installed following FAA STC No. SA1377SO. | 18.04.2007 |
| 10014368 | AERODISA S.L. | EASA.A.S.01638 | | | EMBRAER EMB-120 | Structural Permanent element for a Winch installation. | | 1. Doc. AED2005-02M_MDL #Master Data List#, Ed. 3 Rev. 0, dated 5 October 2008.2. Doc. AED2005-02M_D02 #Aircraft Maintenance Manual Supplement#, Ed. 3 Rev. 0, dated 5 October 2008.3. Doc. AED2005-02M_SW D #Aircraft Wiring Diagram Supplement#, Ed. 3 Rev. 0, dated 5 October 2008. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 04.11.2008 |
| 10014475 | AERODISA S.L. | EASA.A.S.02537 | | | FALCON 20 | AED2006-01M: TCAS II Installation for Falcon 20. | | # Master Data List AED2006-01M_MDL Rev.5 dtd. 26.5.07 or later EASA approved revision.# FM Supplement AED2006-01M_D03 Rev.2 dtd. 9.5.07 or later EASA approved revision. | 1. See FM Supplement AED2006-01M_D03 Section I: LIMITATIONS.2. Certification basis CS 25 Orig issue.3. TGL 8 Rev.2 "ACAS II" complied with.4. TGL 13 Rev.1 "Elementary Surveillance" complied with.5. Prior to installation of this modifi (...) | 18.07.2007 |
| 10014608 | AERODISA S.L. | EASA.A.S.02734 | | | EMBRAER EMB-120 | AED2006-03M: TAWS Installation on EMBRAER 120. | | # Master Data List AED2006-03M_MDL Rev.6 dtd. 25 June 07 or later EASA approved revision.# FM Supplement AED2006-03M_MDL_D03 Rev.4 dtd. 25 June 07 or later EASA approved revision. | 1. See FM Supplement AED2006-01M_MDL_D03 Section II: LIMITATIONS.2. Certification basis per ANAC TCDS EA-8505-RBHA 25 corresponding to FAR 25 Amrt 54.3. TGL 12 complied with.4. SANDEL mandatory SB 3400-09 to be incorporated.5. Prior (...) | 05.07.2007 |
| 10014609 | AERODISA S.L. | EASA.A.S.02735 | | | FAIRCHILD SWEARINGEN SA 228 SE FAIRCHILD SWEARINGEN SA 227 SE | AED2006-04M: TAWS installation on Fairchild SA 228/227 | | Master Data List AED2006-04M_MDL Revision 3, dated 19/04/2007 or later EASA approved revision.FM Supplement AED2006-04M_D03, Revision 1, dated 22/03/2007 or later EASA approved revision. | See FM Supplement AED2006-04M_D03 Section II: Limitations.Certification basis per FAA TCDS A8SW: FAR 23 Amrt 6.TGL 12 complied with. | 31.05.2007 |
| 10014735 | AERODISA S.L. | EASA.A.S.02918 | | | MYSTERE FALCON 20 | AED2006-14M: TAWS installation for Falcon 20. | | Master Data List AED2006-14M_MDL Ed.00, Rev. 3, dated 22-07-2007 or later EASA approved revisions.FM Supplement AED2006-14M_D03 Ed. 00,Rev. 2, dated 12-07-2007 or later EASA approved revisions. | See FM Supplement AED2006-14M_D03 Section II "Limitations".Certification basis: CS25 initial issue.TGL 12 complied with.SANDEL mandatory SB 3400-09 to be incorporated. | 26.07.2007 |
| 10014748 | AERODISA S.L. | EASA.A.S.02933 | | | MD80 SERIES | LOPA for MD80 Series 168 Pax/ Layout passenger arrangement to allow 168 passengers capacity. | | Master Data List AED 2006-12M_MDL, Ed. 1, Rev. Original, dated 10th May 2007.Maintenance Manual Supplement AED 2006-12M_MMS, Ed. 1, Rev. Original, dated 10th May 2007.Installation Manual AED 2006-12M_DIM, Ed. 1, Rev. Original, dated 10th May 2007. | The installation should not be incorporated in any other aircraft unless it is determined that the interrelationship between this installation and any previously approved configuration will not introduce any adverse effect upon the airworthiness of the aircraft. | 31.05.2007 |
| 10014776 | AERODISA S.L. | EASA.A.S.02974 | | | PIPER PA-23-250 | Upgrade of NAV/COM Equipment for IFR navigation. | | Pilot's operating handbook supplement No.14, 15. | SN: 27-7754134. | 18.07.2007 |
| 10015273 | AERODISA S.L. | EASA.A.S.03756 | | | FALCON 900 | Upgrade of dual Mode S SSR Transponder installation to provide both Elementary Mode S Surveillance Flight ID and Enhanced Mode S Surveillance capability (EHS Capable). | | See Aerodisa documents:# Master Data List AED2008-09_F900_EHS_MDL Revision 04# Flight Manual Supplement AED2008-09_F900_EHS_D03 Revision 03# Maintenance Supplement Manual AED2008-09_F900_EHS_D02 Revision 01# Installation Manual AED2008-09 (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pro (...) | 27.03.2009 |
| 10028207 | AERODISA S.L. | P-EASA.A.S.03152 | REV. 1 | | FALCON 2000 | Pocket Door Installation for Falcon 2000 in accordance with Aerodisa Master Data List AED2007-09M_MDL Revision 3, dated 04th December 2009 or later revisions approved by EASA | EASA CS25 amendment 1 plus EASA Certification Review Item (CRI) D-01 Issue 02 "Installation of Interior Door" for those areas, systems, parts and appliances of the original Falcon 2000 design affected by this design change. The certificated noise and/ o (...) | - Aerodisa Master Data List AED2007-09M_MDL Revision 3, dated 04th December 2009.- Aeroplane Flight Manual Supplement, AED2007-09M_AFMS Revision 01, dated 24th June 2009.- Weight and Balance Supplement AED2007-09M_W<(>&c)>B Revision 00, dated 03rd April (...) | Design Change limited to Falcon 2000 aircraft serial numbers 55 | 21.12.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|----------------------|--|--|--|--|--|---------------|
| 10030750 | AERODISA S.L. | P-EASA.A.S.03802 | | | A310 | EHS upgrade on Honeywell TRA-67A066-01127-1402 This STC upgrades existing transponder system by installing Mode S/Enhance surveillance (EHS). | | -, Master Data List, ref. AED2008-10M_MDL Revision 4, dated 19 May 2009 (or any later approved revision) -, AFM Supplement, ref. AED2008-10M_D03, dated 20 December 2008 | None. | 03.07.2009 |
| 10031538 | AERODISA S.L. | | | | DC9-80 (MD80) | Aeronautical Capsat Transceiver | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. EASA CRI H-001/SC.H-001 | -AED2009-24M_AMMS Maintenance Manual Supplement-AED2009-24M_MDL Master Data List-AED2009-24M_MDLO Master Data List for Operators-AED2009-24M_W<->-&->B Weight & Balance-AED2009-24M_DIM Installation Manual-AED2009-24M_AWDS Wiring Diagram Supplementor (...) | None | 24.08.2010 |
| 10040651 | AERODISA S.L. | | | | B300 | FMS Upgrade for King Air B300. | The Certification Basis for the original product remains applicable to this certificate/ approval. | Technical Documents to be Approved by EASA: AERODISA Master Data List AED2011-17M_MDL ED/Rev 0/for later revisions of the above listed documents approved by EASA. Other: AERODISA AED2011-17M Airplane Flight Manual Supplement | S/N Limitation: AIRPLANE MODEL HAWKER BEECHCRAFT B300 S/N FL-255 | 16.07.2012 |
| 10041087 | AERODISA S.L. | | | | DC 9-87, MD88 DC9-81, DC9-82, DC9-83, | Equipment update for MD80 series SatCom System - Update for MD80 Series SatCom System approved under EASA STC | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-001 The requirements for environmental protection and the associated certificated n (...) | Aerodisa Doc: AED2011_14_MDL Master Data List AED2011_14M_AFMS Aircraft Flight Manual Supplement or later revisions of the above listed documents approved by EASA. | This STC is applicable only to aeroplanes that have already incorporated Aerodisa STC certified as EASA STC 10031538 dated 24 August 2010 | 20.08.2012 |
| 10016341 | AERODYME CORPORATION | EASA.IM.A.S.02654 | | FAA STC SA00246BO | COMMANDER AIRCRAFT 114 COMMANDER AIRCRAFT 114A | Installation of Lycoming IO-540-T4B5 engine. | | FAA approved Aerodyme IO-540 engine Supersure Installation Instructions, Doc. No. AD725-01, dated 02 November 2006 or later approved version. FAA approved Aerodyme IO-540 Engine Supersure Instructions for Continued Airworthiness, Doc. No. AD726-01 (...) | | 27.11.2007 |
| 10016701 | AERODYME CORPORATION | EASA.IM.E.S.01016 | | FAA STC SE00242BO | LYCOMING ENGINES IO-320-B | Installation of induction intake pipes in accordance with Aerodyme 114/A/B/TC 320 HP Engine Conversion Installation Instructions, dated July 20th 2006, or later revisions of the above listed document approved by EASA in accordance with ED decision 20 (...) | | Instructions for operation, maintenance and inspection: Doc.No. AD526-01: 114/A/B/TC 320HP Engine Conversion # Instructions for Continued Airworthiness | 1. This STC must be installed concurrently with STC EASA.IM.A.S.02653.2. If the holder agrees to permit another person to use this certificate to alter a product, the holder must give the other person written evidence of that permission. | 04.03.2008 |
| 10034211 | AERO-DYNE REPAIR SERVICES LTD. | | REV. 1 | TCCA STC SH08-41 | BELL 407 | External MGT and Torque Indicating System Installation | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -FMS403101 revision IR dated 4 April 2008 - Flight Manual Supplement; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision)-EO403101 revision 1 dated (...) | None | 02.08.2011 |
| 10038920 | AEROFDC WA LP | | REV. 1 | FAA STC SR02059SE | EC135 P2+, EC135 P2 | Installation of an engine IBF system in accordance with the FDC/aerofilter Master Document List 1135-SERIES-MDL-5 and FDC/aerofilter Installation Instructions 1135-Series-INS-5. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | For model EC135 P2+.. Flight Manual Supplement 1135-RFMS-1.Rev. - For model EC135 P2-, Flight Manual Supplement 1135-RFMS-1, Rev. A or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subs (...) | None | 27.08.2012 |
| 10040685 | AEROFDC WA LP | EASA.IM.R.S.00273 | REV. 1 | FAA STC SR00811SE | AS 350 B, AS 350 B1, AS 350 B2 AS 350 B3, AS 350 BA | Installation of engine inlet air filtration system jaw Master Drawing List No 1350-1001 revision D and Installation Instructions 1350-INST-1, revision IR Notes: 1. STC EASA.IM.R.S.00273 issued on 28 October 2004 was the validation of FAA STC SR00811SE fi (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Design and Installation documentation: Master Drawing List No 1350-1001 revision D dtd November 15th, 2000 Installation Instructions 1350-INST-1, revision IR dtd May 1, 2000 Maintenance documentation: Document 1350-SERIES-ICA-1, revision IR dtd March 1, (...) | This STC is not applicable to AS350B3 model helicopters incorporating modification OP-3369 (2370kg weight extension) This STC is not applicable to AS350B2 model helicopters equipped with VEMD major modification This STC is not applicable to AS350B3 model (...) | 19.07.2012 |
| 10016764 | AEROLITE GmbH | EASA.IM.R.S.01095 | | | EUROCOPTER DEUTSCHLAND MBB-BK | Installation of EC 145 EMS Equipment P/N 145121-501 | | Flight Manual Supplement FMS-506 | 1. No Smoking when Medical Oxygen Cabinet is installed. 2. The LOX System can only be operated during flight when appropriately trained medical crew are on board. 3. This STC is approved only for the product configuration as defined in the approved de (...) | 05.09.2005 |
| 10017183 | AEROLITE GmbH | EASA.R.S.01316 | | | EUROCOPTER DEUTSCHLAND BK117 C | Installation of NATO Stretcher Retainer. | | - Flight Manual Supplement FMS-06-008, Rev. N/C, dated 27.09.2006- Instructions for Continued Airworthiness and Operation Manual OPM-06-010, dated 04.10.2006. | The NATO Stretcher (NSN 6530-01-380-7309) referred to in this installation is not part of this STC. | 03.11.2006 |
| 10017241 | AEROLITE GmbH | EASA.R.S.01429 | REV. 3 | | AB139, AW139 | EMS Kit Light Lateral Stretcher P/N 139127-501 This change extends the compatibility of the installation of configuration 1 of EMS Kit Light Lateral Stretcher P/N 139127-501, approved under EASA STC no. 10017241, Rev. 2 to the AW/AB139 two parts floor. | JAR-29 Amdt. 3 The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement: FMS-07-011 Rev. C dated 20 March 2013 or later revisions of the above listed documents approved by EASA Master Document List MDL-07-011 Rev. C. Instruction for Continuous Airworthiness and Operation Manual: OPM-07-013 Rev. C dated 20 March 2013 | The installation of this modification by third persons is subject to written permission of the approval holder. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previous (...) | 29.05.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------|----------------|----------------------|------------------------------|--|--|--|---|---------------|
| 10017280 | AEROLITE GmbH | EASA.R.S.01528 | REV. 4 | | MBB-BK 117 C-2 | Stylene Package (VIP Kit) Aerolite P/N 145305-501/-502 (Revision to EASA.R.S.01528) Passenger Seat Modification P/N 021376-502, Curtain Inset P/N 145319-502 and Finishing P/N 145302-506/-507 | FAR 29 The Certification Basis for the original product remains applicable to this certificate/ approval | Rotorcraft Flight Manual FMS-08-009 Revision B or later revisions of the above listed documents approved by EASA Operation Manual OPM-08-011 Revision C Master Document List MDL-08-017 Revision E | When the Cargo Separation Wall is installed the original BK117 C-2 smokedetector kit must be installed in the cargo compartment. When operating the rotorcraft with the doors open or removed, the limitations given in Flight Manual Supplement No. FMS 9 (...) | 02.03.2012 |
| 10033323 | AEROLITE GmbH | | REV. 1 | | 429 | EMS Equipment P/N 429020-501/-502 This STC revision introduces the new EMS Equipment P/N 429020-502. In addition the following modifications to the existing EMS Equipment P/N 429020-501 are covered by this change:- Introduction of Cargo Platform P/N 42 (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | For EMS Equipment P/N 429020-501:- Flight Manual Supplement, FMS-429020-501, Rev. A- Airworthiness Limitations, AWL-429020-501, Rev. A, dated 19.02.2011 For EMS Equipment P/N 429020-502:- Flight Manual Supplement, FMS-429020-502, Rev. 0- Airworthin (...) | None | 24.05.2011 |
| 10036324 | AEROLITE GmbH | | REV. 1 | | AB 139, AW 139 | EMS Equipment P/N 139250-501 Rev. 1 introduces a redesign of Base Plate P/N 139252-501 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement FMS-139250-501 Rev A, dated June 28, 2013- Airworthiness Limitation AWL-139250-501 Revision N/Cor later revisions of the above listed documents approved by EASA- Master Document List MDL 10-009 Revision B- Master Drawing List DWL 10-009 Revision B- Equipment List EQL 10-00 | None | 19.07.2013 |
| 10037576 | AEROLITE GmbH | | REV. 2 | | AW109SP | EMS Equipment P/N 109-0002060 This major change to EMS Equipment P/N 109-0002060, approved under EASASTC No. 10037576, Rev. 1, covers the introduction of the Swivelling Adapter P/N 001-0019851-503 as superseding alternative to Swivelling Adapter P/N 00 (...) | CS-27, Initial Issue The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement: FMS-109-0002060, rev A dated 10 April 2013; or later revisions of the above listed documents approved by EASA. Instruction for Continued Airworthiness and Operation Manual: OPM-109-0002060-XXX, rev C dated 2 May 2013. | Swivelling Adapter P/N 001015-503 approved under EASA STC No. 10037576, Rev. 1 are not eligible for new installations. Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherp (...) | 17.05.2013 |
| 10039929 | AEROLITE GmbH | | REV. 1 | | AW109SP | EMS Equipment P/N 109-000862-501 The medical interior is of modular design to allow up to three different EMS stretcher cabin configurations and working positions for the rescue personnel. The EMS Equipment is based on a foldable stretcher mounted onto (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AWL-109-000862-501 Rev. N/C dated 31 May 2012 FMS- 109-000862-501 Rev. A or later revisions of the above listed documents approved by EASA. DWL-12-001 Rev. A dated 24 January 2013, CPL-11-005 Rev. A dated 21 January 2013 | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce adverse effect upon the airworthiness of the product (...) | 18.02.2013 |
| 10042079 | AEROLITE GmbH | | | | AW139 | *EMS Equipment P/N 139-0004728-501*. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FMS-139-0004728-501 Flight Manual Supplement (FMS) for the EMS Equipment P/N 139-0004728-501; AWL-139-0004728-501 Airworthiness Limitations (AWL) for the EMS Equipment P/N 139-0004728-501; or later revisions of the above listed documents approved by EASA. | None. | 06.11.2012 |
| 10043038 | AEROLITE GmbH | P-EASA.R.S.01241 | REV. 1 | | EK117 C-2 | Addition / Integration of EMS Equipment 145121-502 into existing STCEASA.R.S.01241 | | Flight Manual Supplement FMS-506, Rev. C, dated 07.09.2006 Operation Manual OPM-506, Rev. C or later for p/n 145121-501 Operation Manual OPM-06-011, Rev. n/c or later for p/n 145121-502 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined (...) | 29.01.2007 |
| 10014816 | AEROLITE MAX BUCHER AG | EASA.A.S.03024 | | | BOMBARDIER CL-600-2B16 (604) | 3rd Stretcher and Cargo Platform. P/N 604489-501 | | Flight Manual Supplement: FMS-07-001 Instructions for continued airworthiness and operations manual: OPM-07-002 | S/N: 5529/5530/5540 | 20.06.2007 |
| 10014983 | AEROLITE MAX BUCHER AG | EASA.A.S.03275 | | | MD-11 | Modification of two SICMA Mono F/C Seat Type 17011(X)1-(XX) as detailed in Lufthansa Supplemental Type Definition Document STDD 25-05/32 R02 and in accordance with Aerolite Bucher Master Document List MDL-06-015 Rev. B for installation in MD-11 MSN48533. | | MDL-06-015 Rev. B Master Document List MDL-07-006 Installation Manual VIP Seat installation. | It is the responsibility of the installer to make sure that this installation is certified for the use in effective aircraft. Compliance with the following Cabin Safety requirements is not covered by this approval: 1. FAR/JAR 25.791 Passenger Inform (...) | 16.11.2007 |
| 10015764 | AEROLITE MAX BUCHER AG | EASA.IM.A.S.01800 | REV. 2 | | L35A | EMS Equipment Installation Introduction of stretcher configuration P/N Aerolite 250071-506 and power box installation P/N Aerolite 250080-503 into existing EMS configuration P/N Aerolite 250070-501 | FAR 25 Amdt. up to Amdt. 25-18 for aircraft cert. basis The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: SCS or SDFs: according EASA.IM.A. (...) | Master document list MDL-06-001 Flight Manual Supplement FMS-06-001 Operations Manual OPM-06-001 or later revisions of the above listed documents approved by EASA | Limited to: S/N: 628, 663 | 06.01.2010 |
| 10016720 | AEROLITE MAX BUCHER AG | EASA.IM.R.S.01014 | | | AGUSTA A-109 E | Installation of Medical Oxygen Cabinet P/N 109111-501 | | -Installation Manual IM-502 | 1. No Smoking when Medical Oxygen Cabinet is installed. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.05.2005 |
| 10016722 | AEROLITE MAX BUCHER AG | EASA.IM.R.S.01019 | | | MD HELICOPTERS MD900 | External Hoist Mirror Installation P/N 9005024-501, P/N 9005024-502, P/N 9005024-503, P/N 9005024-504. Flight Manual Supplement FMS-404. | | Flight Manual Supplement FMS-404. | 1. Only for VFR operations. 2. Only approved configurations to be installed. 3. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft (...) | 08.07.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|----------------------------|----------------|----------------------|-------------------------------------|--|--|---|--|---------------|
| 10016742 | AEROLITE MAX BUCHER AG | EASA.IM.R.S.01068 REVISION | REV. 1 | | AGUSTA AB-139 | Installation of Aerolite EMS equipment P/N 139001-501, 139001-502 | | Rotorcraft Flight Manual Supplement FMS-502, FMS-510 Operational Maintenance Manual OPM-501MDL-405 rev A | 1. No smoking permitted when the Oxygen system is installed.2. The occupant of the LH forward Aft facing seat must wear a helmet.3. For the installation of the EMS equipment the following Agusta equipment must be installed:-the EMS Electrical (...) | 19.06.2006 |
| 10016765 | AEROLITE MAX BUCHER AG | EASA.IM.R.S.01096 | REV. 1 | | MBB-BK 117 C-2 | EMS Equipment P/N 145146-501 Revision 1 introduces a new stretcher and various provisions for installation of medical equipments. | FAR 29The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FMS-505 rev A or later approved revisions of the above listed documents approved by EASA OPM-504 rev AMDL-504 rev A | None | 15.04.2011 |
| 10016793 | AEROLITE MAX BUCHER AG | EASA.IM.R.S.01143 | REV. 1 | | EUROCOPTER DEUTSCHLAND MBB-BK | EMS Equipment P/N 145181-501 Introduction of a second configuration with a secondary RH stretcher installation on EMS integral floor | | Flight manual supplement FMS-507- Master Document list MDL-507 Revision A | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer.2. No smoking in the rotorcraft | 22.06.2006 |
| 10016794 | AEROLITE MAX BUCHER AG | EASA.IM.R.S.01144 | REV. 1 | | AGUSTA A109S | A109S EMS equipment P/N 109115-50X. This revision 1 of EASASTCEASA.IM.R.S.01144 is related to the introduction of stretcher configuration Aerolite P/N 1090117-503. | As defined in EASA TCDS R.005.The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement - FMS-508 Revision C Operation and Maintenance Manual - OPM-508 Revision BMDL-508 Revision B or later revisions of the above listed documents approved by EASA | The Agusta provision for EMS kits (Aerolite) P/N 109-0814-22-501 has to be installed. | 18.01.2010 |
| 10016817 | AEROLITE MAX BUCHER AG | P- EASA.IM.R.S.01182 | REV. 2 | | MBB-BK 117 C-2 | EMS Equipment P/N 145212-501 This change introduces the following changes to EMS Equipment P/N 145212-501 (EASA approved under STC No. 10016817, Rev. 1):(1) Modification of configuration no. 2(2) Introduction of derivate configurations no. 3, 4, 5 and (...) | FAR 29 Amdt: 40The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement: FMS-06-002 at Rev B Instructions for Continued Airworthiness: OPM-06-002 Rev D or later revisions of the above listed documents approved by EASA. | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 05.04.2013 |
| 10016820 | AEROLITE MAX BUCHER AG | EASA.IM.R.S.01185 | | | EUROCOPTER DEUTSCHLAND EC-135 | EMS Equipment P/N 135138-502 This major modification covers an EMS equipment including medical seats, stretcher, oxygen cylinder rack, stowage cabinet and Oxylog and Propaq retainers Some components, identified in the equipment list EQL-06-003 and air (...) | | Flight manual supplement FMS-06-003 | 1. Limited to S/N 01052. No Smoking when Medical Oxygen cabinet is installed.3. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft (...) | 22.03.2006 |
| 10017085 | AEROLITE MAX BUCHER AG | EASA.R.S.00397 | REV. 2 | | EC135 SERIES | EMS Equipment DRF - P/N 135405-501/-502. Revision 1 to extend approval from P1/P2, T1/T2 to all EC135 models. Revision 2 to introduce reversible C/P seat installation configuration P/N 135331-503 into EMS equipment configuration P/N 135405-501 and toin (...) | JAR 27 Issue 1The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Flight Manual Supplement FMS-202 Revision E (valid for P/N 135405-501)-Flight Manual Supplement FMS-410 Revision C (valid for P/N 135405-502) or later revisions of the above listed documents approved by EASA-Operation Manual OPM-202 Revision D-Master Document List MDL-201 Revision F | P/N 135301-503 and P/N 135543-503 are only valid for the rotorcraft S/N 870 and subsequent. | 05.02.2010 |
| 10017088 | AEROLITE MAX BUCHER AG | EASA.R.S.0057 | REV. 3 | | EC 135 | EMS Equipment P/N 135375-501 / 135375-502 STC revision 3 relates to - EMS Integral floor Inst. P/N 135301-503 incl. EO-10 017 (Aeration Cover Kit P/N 0071664-501, applicable to EMS Equipment P/N 135375-502 only)- Med. Equipment Rails Inst. P/N 135552-5 (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and (...) | For configuration P/N 135375-501: Flight Manual Supplement FMS-002 Rev. J Chapter 05 of Operations Manual OPM-002 Rev D, dated July 20, 2012 For configuration P/N 135375-502: Flight Manual Supplement FMS-412 Rev. E Chapter 05 of Operations Manual OPM-002 (...) | none | 28.08.2012 |
| 10017157 | AEROLITE MAX BUCHER AG | EASA.R.S.01265 | | | EUROCOPTER DEUTSCHLAND MBB-BK | APC-06-006 EMS equipment installation fro BK-117-C2 (EC145), part number 145221-501 | | Flight manual supplement FMS-06-005 Operation and maintenance manual OPM-06-007 | | 24.07.2006 |
| 10017164 | AEROLITE MAX BUCHER AG | EASA.R.S.01272 | | | EUROCOPTER DEUTSCHLAND EC-135 | Installation of EMS equipment Aerolite P/N 135570-501 | | Flight Manual Supplement FMS-06-004, Issue N/C dated 22.05.2006 or later EASA approved revision Operations Manual OPM-06-006 Installation per Drawing List DWL-06-005 | Only S/N 0041 | 15.06.2006 |
| 10017172 | AEROLITE MAX BUCHER AG | EASA.R.S.01286 | REV. 1 | | EUROCOPTER DEUTSCHLAND EC-135 | Installation of EMS Equipment P/N 135500-501 or P/N 135500-502. This revision extends the applicability to EC135P2+ and EC135T2+. | | Flight Manual Supplement FMS-210, Revision D, dated 26 March 2008 (for EMS equipment P/N 135500-501, EC135 up to and including S/N 0336) Flight Manual Supplement FMS-407, Revision B, dated 26 March 2008 (for EMS equipment P/N 135500-502, EC135 S/N 0337 and upwards). | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 29.05.2008 |
| 10017187 | AEROLITE MAX BUCHER AG | EASA.R.S.01324 | REV. 2 | | EC135 P1, P2, P2+ EC135 T1, T2, T2+ | EMS Equipment P/N 135572-501 This change to the STC 10017187, rev 1 covers:- The introduction of improved Seat AFT RH Inst. P/N 135-0022016-502 instead of Floor Plate AFT RH Inst. P/N 135534-502/-504/-506/-508 for EMS config. no. 1 and 2.- The introduc (...) | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement: FMS-06-007 rev D dated 23 November 2012 or later revisions of the above listed documents approved by EASA Instruction for Continued Airworthiness: OPM-06-009 rev D dated 23 November 2012. | S/N: 337 and up The TC holder optional Air Conditioning System is not compatible with this installation up to and including S/N 869. New installations of this STC are only allowed with the revision covered by this approval or later approved revisions. P (...) | 07.02.2013 |
| 10017194 | AEROLITE MAX BUCHER AG | EASA.R.S.01331 | | | AGUSTA AB-139 AGUSTA AW-139 | EMS Medical Rack Installation p/n 139099-501 | | Reference Master Document List MDL-06-024 Flight Manual Supplement FMS-06-012 Operational Manual OPM-06-014 | EMS (Medevac) electrical provisions 3G2564A01441 must be installed with modifications according to 3G2564W00201 (page 1 only). | 08.03.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|----------------|----------------|----------------------|--|--|---|--|--|---------------|
| 10017209 | AEROLITE MAX BUCHER AG | EASA.R.S.01361 | | | AGUSTA AB-139 AGUSTA AW-139 | P/N 139086-501. The equipment contains a specific cabin floor layer, 2 medical seats installed in the aft part of the cabin, one or 2 stretchers, a medical supply cabinet and various storage compartments. | | 1. Flight Manual Supplement: FMS-07-0062. ICA and Operations manual: OPM-07-006. | 1. When oxygen is installed No smoking permitted.2. The occupant of the LH FWD Alt facing seat must wear a helmet.3. When the secondary stretcher is occupied the head protection pad must be installed between the stretchers.4. Following e (...) | 21.11.2007 |
| 10017214 | AEROLITE MAX BUCHER AG | EASA.R.S.01371 | | | EUROCOPTER AS365 SERIES EUROCOPTER DEUTSCHLAND EC-135 EUROCOPTER DEUTSCHLAND EC-135- EUROCOPTER DEUTSCHLAND EC-635 | Armasuisse 1 LH Stretcher Installation. P/N 135586-501. | | # Flight Manual Supplement FMS-07-002.# Operational manual OPM-07-003.# Installation Manual IM-07-001. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 03.07.2007 |
| 10017215 | AEROLITE MAX BUCHER AG | EASA.R.S.01372 | | | EUROCOPTER DEUTSCHLAND EC-135- EUROCOPTER DEUTSCHLAND EC-635 EUROCOPTER DEUTSCHLAND EC-635- | Armasuisse 2 RH Stretcher Installation P/N 135587-501. | | # Flight Manual Supplement FMS-07-003.# Operational manual OPM-07-004.# Installation Manual IM-07-002. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 03.07.2007 |
| 10017216 | AEROLITE MAX BUCHER AG | EASA.R.S.01373 | | | EUROCOPTER DEUTSCHLAND EC-135 EUROCOPTER DEUTSCHLAND EC-635 | Life Support Module installation. P/N 135588-501. The installation consists of a storage provision for up to three oxygen bottles. The is module installed in the forward part of the passenger cabin. | | Flight manual supplement: FMS-07-004. Instruction for continued airworthiness and operations manual: OPM-07-005. | | 31.10.2007 |
| 10017217 | AEROLITE MAX BUCHER AG | EASA.R.S.01374 | REV. 2 | | EC 135 P1 (CDS) EC 135 P1 (CPDS) EC 135 P2 (CPDS) EC 135 P2+ EC 135 T1 (CDS) EC 135 T1 (CPDS) EC 135 T2 (CPDS) EC 135 T2+ | EMS Stretcher Inst. P/N 135525-501/502/503/504. This STC 10017217 Rev. 2 is revision of STC No. EASA.R.S.01374 Rev. 1, introducing the following components/modification: - Modification of the pre-requisite for the Secondary Carbon Stretcher Installation P/ (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement: FMS 06-006 rev B or later revisions of the above listed documents approved by EASA. Instruction for Continued Airworthiness: OPM-06-008 rev B. Installation Instructions: IM-06-009 rev B | S/N Installations P/N 135525-501 to -504 are valid for installation on the Aerolite Integral Floor Inst. P/N 135301-50X with the following limitations: P/N 135525-501 and -503 are eligible for installation on all EC135 models up to and including S/N 3 (...) | 13.03.2013 |
| 10017229 | AEROLITE MAX BUCHER AG | EASA.R.S.01400 | | | EUROCOPTER DEUTSCHLAND BK117 C | EMS Stretcher Installation Police Thüringen (Aerolite P/N 145285-501). | | Flight Manual Supplement FMS-07-009, Rev. N/C, dated 1 September 2008. Operational Manual OPM-07-011 | | 12.11.2008 |
| 10017231 | AEROLITE MAX BUCHER AG | EASA.R.S.01403 | REV. 1 | | AGUSTA AB-139 AGUSTA AW-139 | Installation P/N 139084-501 for 3 stretcher units with relevant cabin floor rails and featuring each, dedicated oxygen and electrical distribution systems and oxygen bottles stowage provisions. | | 1. Flight manual supplement: FMS-07-0082. Installation, maintenance and operations according to Aerolite Manual OPM-07-010. | 1. Agusta EMS (Medevac) electrical provisions 3G2564A01411 must be installed with modifications according 3G2564W00201.2. When oxygen is installed No smoking permitted.3. Required minimum crew is defined in FMS-07-008.4. The dimensional limitations for medical rails are defined in OPM-07 | 28.11.2007 |
| 10017232 | AEROLITE MAX BUCHER AG | EASA.R.S.01406 | REV. 1 | | EC 135 T2, EC 135 P2, EC 135 T2+, EC 135 P2+ EC 135, EC 135 T1, EC 135 P1, EC635 P2+ EC635 T1 (CDPS), EC635 T2+ | EMS Stretcher Installation P/N 135591-501. Introduction of the Cirrus Stretcher Model CI-1 01 0-123-01 into the already certified Cirrus 1010 Stretcher Installation P/N 135574-501/-502/-503/-504 (STC EASA.R.S.01406) | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Flight manual supplement FMS-07-007 Revision A or later revisions of the above listed document approved by EASA. Instructions for Continued Airworthiness: OPM-07-009 Revision A Master Document List (for configuration) MDL-07-006 Revision A | To ensure patient head protection from injuries during crash, there must be at least 200mm of space between the forward end of each installed stretcher and any equipment installed forward of the stretchers. | 20.04.2010 |
| 10017249 | AEROLITE MAX BUCHER AG | EASA.R.S.01448 | | | AGUSTA AB-139 AGUSTA AW-139 | EMS Multi-Mission Kit Installation of the EMS Multimission Kit Aerolite-P/N 139124-501 into cabin of Agusta Westland rotorcraft model AB/AW 139. The Kit covers EMS equipment (ref. Aerolite-P/N 139086-501) certified per EASA.R.S.01361 adding a new so-cal (...) | | (R)FMS-08-003 Rev N/C, dated February 28th, 2008. OPM-08-003 Rev. N/C, dated January 23rd, 2008 | Rotorcraft Operational Limitations: (ref. (R)FMS-08-003 Rev N/C)-When oxygen is installed no smoking permitted-When equipment rack is installed occupant of LH FWD aft-facing seat must wear a helmet-When secondary stretcher is occupied head protection pad must be installed | 04.04.2008 |
| 10017252 | AEROLITE MAX BUCHER AG | EASA.R.S.01454 | REV. 2 | | EC135 P1(CDS), EC135 P1(CPDS) EC135 P2(CPDS), EC135 P2+ EC135 T1(CDS), EC135 T1(CPDS) EC135 T2(CPDS), EC135 T2+ | Change to Installation of EMS Equipment Aerolite P/N 135602-502. Change comprises of:- Introduction of Support Panel Assy P/N 004191-501 into DC Power System Inst. P/N 135615-501- Introduction of EMS Integralfloor Inst. P/N 135301-503- Introduction of R (...) | JAR 27 The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement FMS-08-001 Revision B, dated May 6th, 2011 or later revisions of the above listed documents sections approved by EASA Master Document List MDL-08-004 Equipment List EQL-08-004 Operations Manual OPM-08-001 Revision B, dated May 6th, 2011 | This STC is limited to S/N 337 and above. | 15.10.2012 |
| 10017263 | AEROLITE MAX BUCHER AG | EASA.R.S.01474 | REV. 1 | | AGUSTA AB-139 AGUSTA AW-139 | EMS Provisions Kit. Revision of STC for incorporation of PAX seat configurations Aerolite P/N 139117-505; 139117-506 and 139117-507 of 12, 14 and 15 PAX seats | | FMS-08-002 Revision AMDL-08-005 Revision A OPM-08-002 Revision A | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 30.03.2009 |
| 10017265 | AEROLITE MAX BUCHER AG | EASA.R.S.01479 | REV. 2 | | AW139 | AW 139 EMS Kit P/N 139097-501 Revision 1 related changes: Stretcher Inst. P/N 139190-501/-503/-505/-507/-511/-513/-515/-517/-601/-603/-605/-607/-611/-613/-615/-617- Cargo Platform Inst. P/N 139191-501- Toughbook Retainer Inst. P/N 139192-501- Toughbo (...) | JAR 29 Amdt: 3 The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Initial technical documentation: Airworthiness Limitations Section of FM Supplement FMS-08-005 Revision N/C Airworthiness Limitations Section of Operations Manual OPM-08-005 Revision N/C, dated May 19, 2008. Revision 1 related technical documentation: Airworth (...) | None | 08.11.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|------------------|----------------|--------------------------------|--|--|---|--|--|---------------|
| 10017266 | AEROLITE MAX BUCHER AG | EASA.R.S.01481 | | | EUROCOPTER DEUTSCHLAND EC-135 | EC135 EMS Equipment.Update of (R)FMS-401 for EMS equipment P/N 135545-501 with validation for all EC135 models up to S/N 0337.Introduction of new EMS equipment variant P/N 135545-502 with new (R)FMS-08-006 and OPM-08-006 valid for all EC135 models from S/N 0337 on. | | (R)FMS-401 Rev.B or later(R)FMS-08-006 Rev.A or laterOPM-08-006 Rev.A or laterMDL-401 Rev.B or later | Helicopter shall be operated as per limitations section of FMS.Placards as per limitations section of FMS shall be installed.No smoking permitted when Oxygen is installed.Prior to installation of this modification the installer must determine that (...) | 24.06.2008 |
| 10017274 | AEROLITE MAX BUCHER AG | EASA.R.S.01506 | | | EUROCOPTER DEUTSCHLAND BK117 C | Installation of EMS equipment P/N 145268-501 including: 3 medical seats, a primary patient system, medical supply unit,bottle rack, medical storage cabinet, EMS electrical installation, medical equipment retainers, tie down web and stretcher storage provisions. | | 1. Flight manual supplement FMS-08-007 issued 1st September2008 or later approved version.2. OPM-08-008 dated 8 September 2008 or later approved version. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 14.10.2008 |
| 10017278 | AEROLITE MAX BUCHER AG | EASA.R.S.01524 | REV. 2 | | AB139, AW139 | EMS Kit P/N 139172-501/139172-502Revision 1 of STC to incorporate PAX seat configurations Aerolite P/N139117-505, 139117-506 and 139117-507 of 12, 14 and 15 PAX seats (as perSTC EASA.R.S.01524 Revision 1)Revision 2 of STC to introduce Ceiling Unit A (...) | JAR 29 Amdt. 3The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Flight Manual Supplement FMS-08-008 Rev. B or later revisions of the above document approved by EASA-Operations Manual OPM-08-010 Rev. B-Master Document List MDL-08-015 Rev. B | None | 05.02.2010 |
| 10017286 | AEROLITE MAX BUCHER AG | EASA.R.S.01535 | REV. 1 | FOCA STC 25-20-91; EASA.R.S.01 | EUROCOPTER DEUTSCHLAND BK117 C | Installation of EMS equipment P/N 145030-501Note:This STC is based on the FOCA STC 25-20-91 and consists of a stretcher platform, 3 medical seats, 2 medical cabinets and a medical supply unit.This revision 1 of the STC supersedes originalissue of EASA.R.S.01535. | | FMS-216 at revision C.OPM-216 at revision C. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 08.04.2009 |
| 10017292 | AEROLITE MAX BUCHER AG | EASA.R.S.01554 | | FOCA STC 25-20-65 | EUROCOPTER DEUTSCHLAND EC-135 | Clamshell Door Lock installation P/N 135065-501/-502.Note: This STC is based on the FOCA STC Z 25-20-65. Its applicability is extended to P2+ and T2+ models. | | Installation Manual Doc No. IM-901Maintenance Manual Doc. No. 52-31-01 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 23.02.2009 |
| 10026923 | AEROLITE MAX BUCHER AG | P-EASA.R.S.01571 | REV. 3 | | EC135 | EMS P/N 135652-502Design changes covered:- Cargo Platform Installation P/N 135746-501- Storage Compartment Installation P/N 135663-504- Spineboard Retainer Installation P/N 135730-502- Propan Encore Retainer Installation P/N 135747-501- Snorkel Kit P/N 0071559-501 (Engineering Order EO-10-009) | JAR 27 Issue 1The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Related to the actual change: FMS-09-004 Revision B OPM-09-008 Revision B or later revisions of the above listed documents approved by EASAEQL-09-008 MDL-09-008 | S/N: 337 and onwards only | 14.12.2010 |
| 10029101 | AEROLITE MAX BUCHER AG | P-EASA.A.S.03979 | REV. 1 | | BEECH 200 SUPER KING AIR | Change to B200 Medical Interior Kit/P/N 230001-501Change compromises of:- Cargo Platform Inst. P/N 230006-502- StretcherSeat Inst. P/N 230015-502- AC Power Kit P/N 0071605-501- Engineering Orders EO-10-010/-011/-037In addition this Major Change covers (...) | FAR 23The Certification Basis for the original product remains applicable to this certificate/ approval.The certificatednoise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement FMS-09-005 Revision ACh. 4 of Operations Manual OPM-09-007 Revision NC (ALS is unchanged)or later revisions of the above listed documents approved by EASA | Applicable to Beechcraft model B200 S/N BB-2001 and subsequent | 25.02.2011 |
| 10029681 | AEROLITE MAX BUCHER AG | | REV. 2 | | EC135P1(CDS), EC135P1(CPDS), EC135P2(CPDS), EC135P2+ EC135T1(CDS), EC135T1(CPDS), EC135T2(CPDS), EC135T2+ | Utility Missions Interior P/N 135736-502This major change to EASA STC No. 10029681 dated 12.04.2010 covers theintroduction/ modification of the following components into the UtilityMissions Interior P/N 135736-502:- Introduction of the Seat FWD LH. (...) | JAR 27 Arndt: 1st issueThe Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement: FMS-135736-502 rev N/C dated 24 April 2012;Airworthiness Limitation: OPM-135736-502 rev N/C dated 26 April 2012;or later revisions of the above listed documents approved by EASAMaster Document List: MDL-10-002 revA dated 18 July 2012; | None. | 08.08.2012 |
| 10029915 | AEROLITE MAX BUCHER AG | | | | EC135 P1(CDS), EC135 P1(CPDS), EC135 P2(CPDS), EC135 P2+, EC135 T1(CDS), EC135 T1(CPDS), EC135 T2(CPDS), EC135 T2+ | EMS Equipment P/N 135710-502 | JAR 27 Issue 01 i.a.w. TCDS EASA.R.009The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Flight Manual Supplement FMS-10-001 Rev. N/Cor later revisions of the above listed documents approved by EASA-Operations Manual OPM-10-001 Rev N/C-Master Document List MDL-10-001Rev. N/C | Limited to: S/N 870 and subsequent | 03.05.2010 |
| 10030822 | AEROLITE MAX BUCHER AG | P-EASA.R.S.01551 | REV. 3 | | AW/109SP | EMS Kit AW/109SP REGA P/N 1090150-501Changes related to Rev.1:- Introduction of the Propan MD Retainer Inst. P/N 1090170-503 and an additional Training Configuration 2' into the already certified EMSEquipment P/N 1090150-501 (STC EASA.R.S.01551). Mod (...) | The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | FMS-09-001 Rev. D dated 17 September 2013or later revisionsof the above listed documents approved by EASAOPM-09-002 Rev. F dated 23 July 2013 | Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noadverse effect upon the airworthiness of the product. | 24.09.2013 |
| 10030992 | AEROLITE MAX BUCHER AG | | REV. 2 | | MBB-BK 117 C-2 | EMS Equipment P/N 145341-501Change for Rev. 1 of STC comprises of:Addition of Configuration 145341-502(-,-,-)and the following changes compared to the version - 501 of the equipment:- Relocation of a Seat and changes of the Seating configuration. - Re (...) | FAR 29 Amdt. 40 including equivalent safety finding as defined in CRD-1The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | For configuration P/N 145341-501:Flight Manual Supplement FMS-10-003 Rev. AOperations and Installation Manual OPM-10-04Rev. N/Cor configuration P/N 145341-502:Flight Manual Supplement FMS-11-001 Rev. N/COperations and Installation ManualOPM-11-01 R (...) | Compatibility with NVIS has not been demonstrated.Compatibility with TC-holders 3 seat bench has not been demonstrated forconfiguration P/N 145341-501. | 01.08.2011 |
| 10032080 | AEROLITE MAX BUCHER AG | | | | AW139 | EMS Equipment P/N 139200-501 | JAR 29 Amdt. 3The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement FMS-10-004 Revision N/Cor later revisions of the above listed documents approved byEASAOperating Manual OPM-10-005 Rev. N/C/Equipment List EQL-10-004Master Document List MDL-10-004 | Limited to: S/N 41226 onwards | 06.10.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------|----------------|----------------------|---|--|--|---|---|---------------|
| 10033098 | AEROLITE MAX BUCHER AG | | | | EC135 | VIP Equipment P/N 135748-501 | JAR 27 Issue 1The Certification Basis for the original product remains applicable to this certificate/ approvalThe certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | EQL-1 0-01 0 Revision N/CIM-10-004 Revision N/CMDL-10-010 | Reg.: \ , S/N: | 17.12.2010 |
| 10036912 | AEROLITE MAX BUCHER AG | | | | AW139/AB139 | EMS Equipment P/N 139-0003670-501This STC covers 4 EMS cabin configurations and 3 PAX configurations as described in the FMS-139-0003670-501 and according to the Masterdocument list MDL-11-016 rev N/C. | JAR 29 as described in EASA TCDS.R.006The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FMS: FMS-139-0003670-501 rev N/C Airworthiness limitations: AWL-139-0003670-501-0 initial issue or later revisions of the above listed documents approved by EASA Operations manual: OPM-139-0003670-501 rev A | None | 17.10.2011 |
| 10038985 | AEROLITE MAX BUCHER AG | | | | EC135 | EMS Equip. P/N 135-0004171-501The equipment includes up to two stretchers, two seats installed in the forward part of the cabin and stowage between the two seats. | The Certification Basis for the original product remains applicable to this certificate/ approval.The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement: FMS-135-0004171-501-0 rev N/C dated 9.12.2011- Airworthiness Limitation: AWL-135-0004171-501-0 rev N/C dated 6.12.2011 or later revisions of the above listed documents approved by EASA-Master Document List: MDL-11-027 rev N/C dated 22.03.2012 | None | 28.03.2012 |
| 10039944 | AEROLITE MAX BUCHER AG | LBA STC RC 1240 | REV. 1 | | MBB-BK117 C-2 | Change to LBA STC No.RC 1240 on EMS Equipment P/N 145170-501.Addition of variant P/N 145169-501This change introduces:EMS variant P/N 145169-501 composed by:- Introduction of the new Cargo Platform Inst. P/N 145-000835-501-Introduction of the new (...) | The Certification Basis for the original product remains applicable to this certificate/ approval.The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | For EMS Equipment P/N 145170-501: FMS-302 (Flight Manual Supplement)For EMS Equipment P/N 145169-501: FMS-509 (Flight Manual Supplement)or later revisions of the above listed documents approved by EASA.For EMS Equipment P/N 145170-501:OPM-302 (Opera (...)) | Not applicable | 04.06.2012 |
| 10040646 | AEROLITE MAX BUCHER AG | | REV. 1 | | EC 135 P1 EC 135 P2+ EC 135 T1 EC 135 T2 EC 135 T2+ EC 635 P2+ EC 635 T1 EC 635 T2+ | Ballast Platform Installation P/N 135-0012762-501. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement FMS-135-0012762-501, Revision N/C, dated July 6th 2012, or later revisions of the above listed documents approved by EASA - Master Document List MDL-12-010 Revision N/C, dated 12/07/2012;- Weight and Balance manual WBM-135-0012762-501, Initial Issue, dated July 6th, 2012. | None. | 20.07.2012 |
| 10043032 | AEROLITE MAX BUCHER AG | EASA.R.S.00261 | REV. 2 | | EC135 P1/P2, T1/T2 | EMS Equipment P/N135520-501/502/503This STC 10043032 Rev. 2is revision of STC EASA.R.S.00261 Rev. 1, adding cargo platform P/N 135-0000823-501. | The Certification Basis for the original product remains applicable to this certificate/ approval.The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplements:For EMS P/N 135520-501: FMS-303 Revision D, dated January 31, 2013For EMS P/N 135520-502: FMS-408 Revision H, dated January 31, 2013For EMS P/N 135520-503: FMS-501 Revision D, dated January 31, 2013 or later revisions of th (...) | -EMS Equipment P/N 135520-501 is valid for installation on all EC135T1/T2/P1/P2 up to and including S/N 336- EMS Equipment P/N 135520-502 is valid for installation on all EC135T1/T2/P1/P2 from S/N 337- EMS Equipment P/N 135520-503 is valid for instal (...) | 12.03.2013 |
| 10043393 | AEROLITE MAX BUCHER AG | EASA.R.S.01556 | | | AB412 | Quick change EMS Kit | | - (R)FMS-09-003 Rev. NC- OPM-09-005 Initial Issue, dated March 19, 2009- MDL-09-004 Rev. N/C- EQL-09-004 Rev. NC | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The (...) | 24.07.2009 |
| 10044334 | AEROLITE MAX BUCHER AG | | | | B429 | EMS Equipment P/N 429-0011900-501 | CS 27 Amdt: Change 1The Certification Basis for the original product remains applicable to this certificate/ approval.Derogation to latest amendment (Part 21A.101): No significant change.The certified noise and/ or emissions levels of the original (...) | Rotorcraft Flight Manual Supplement FMS-429-0011900-501 Rev NCAirworthiness Limitations AWL- 429-0011900-501 Rev NCInstructions for Continued Airworthiness included in chapter 05 of Operations Manual OPM- 429-0011900-501 Rev A, of Nov 23, 2012 or later (...) | S/N: 57067The medical portable devices, including their operation, have to be approved and compatibility with the helicopter needs to be checked for electromagnetic interference (EMI) and radio frequency interference (RFI) by the operator. Use of these (...) | 05.04.2013 |
| 10046038 | AEROLITE MAX BUCHER AG | | | | 45 | EMS Equipment P/N 250-0023000-501 | JAR 25 Amendment 13 i.a.w. EASA.A.020The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement FMS-250-0023000-501, Rev. 0 of June 28, 2013 Airworthiness Limitations AWL-250-0023000-501, Rev. 0 of July 10, 2013 Operations Manual OPM-250-0023000-501, Rev. 0 of June 30, 2013 or later revisions of the above listed documents (...) | Design approval is valid from and including S/N 45-233 to and including S/N 45-2000 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change (...) | 07.08.2013 |
| 10015544 | AEROMECH INC. | EASA.IM.A.S.01429 | | FAA ST01456SE | BEECH 200 BEECH 200C BEECH 300 BEECH 300LW BEECH B200 BEECH B200C BEECH B300 BEECH B300C | Group approval for reduced vertical separation minimum. | | Aeromech AMI-OD-KAIR-0414, Rev "New", dated 20 AUG 2004.AFMS No. AMI-AFMS-KAIR-0429A, Rev "Original", dated 30 September 2004.AFMS No. AMI-AFMS-KAIR-0429B, Rev "Original", dated 30 September 2004 | 1. AFMS No. AMI-AFMS-KAIR-0429A Rev "Original", dated 30 September 2004 is required on-board for operation with Beech 200, 200C, B200, B200C, 300 and 300LW series.2. AFMS No. AMI-AFMS-KAIR-0429B Rev "Original", dated 30 September 2004 is required on-board for operation with Beech B300, B300C se | 25.11.2005 |
| 10015820 | AEROMECH INC. | EASA.IM.A.S.01892 | | ST01413SE | HAWKER HS 125-700A HAWKER HS 125-700B | Approval of Reduced Vertical Separation Minimum (RVSM). | | In accordance with AeroMech Document No AMI-OD-H700-0010 Rev C, dated December 7, 2005, or later approved revisions. Operation to be in accordance with AeroMech Airplane Flight Manual Supplement ref AMI-AFMS-H700-0417 Rev A, dated, May 16, 2005, or later approved revisions. | 1. Only applicable to Aircraft Serial 257037.2. This STC must be installed after Avionics Mobile STC ref AMI1518. The installation is to be completed by the embodiment of Avionics Mobile Minor change AM1532 installation of RVSM ADMs. | 17.10.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|---------------------|----------------|----------------------|---|---|--|--|--|---------------|
| 10016064 | AEROMECH INC. | EASA.IM.A.S.02243 | REV. 3 | FAA STC ST01526SE | 500 (CITATION I) 550 (CITATION II) 560 (CITATION V) S550 (CITATION S/II) | Group Aircraft Approval for Reduced Vertical Separation Minimum (RVSM) for Cessna Model 500/550/S550/560. This STC is an update of STC EASA.IM.A.S.02243 Rev 1, and incorporates added class of avionics as applicable to the RVSM Group approval in accord (...) | EASA TCDS IM.A.2077The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are (...) | 1) Applicant Technical Documentation (Aeromech) : AMI-QD-CS-0307, Rev F2) Instructions for Initial and Continued Airworthiness: Depending on the class of installed avionics on the aircraft- AMHICA-CS 0509 : for Class 1 airframe (group airframes equ (...) | none | 08.08.2011 |
| 10016065 | AEROMECH INC. | EASA.IM.A.S.02244 | | FAA STC #ST01522SE | CESSNA 500 CESSNA 551 | Validation of FAA STC ST01522SE:Group aircraft approval for RVSM capability. | | Aeromech Document No. AMI-QD-CS-0311, Rev. B, dated 08 Sept. 2006 or later FAA approved revision.FAA approved AeroMech AFM-S Doc. AMI-AFMS-CS-0708, dated 04/04/2007 or later FAA approved revision.FAA approved AeroMech Instructions for ContAW Doc. No (...) | The installer must determine if this design is compatible with previously approved modifications. If the holder agrees to permit another person to use this certification or alter a product, the holder must give the other person written evidence of that permission. | 26.04.2007 |
| 10026757 | AEROMECH INC. | P-EASA.IM.A.S.03086 | | FAA STC SA01602SE | CESSNA 441 SERIES | Group Approval for Reduced Vertical Separation Minimum (RVSM) | FAR Part 23 | Airplane Flight Manual Supplement as per AEROMECH document, "AM0114-FM01 - for Class 1 airframes dated 08 June 2009, or later EASA approved revision: ", "AM0114-FM02 - for Class 2 airframes, dated 08 June 2009, or later EASA approved revision. Instruction (...) | This approval is issued based on the FAA STC SA01602SE, issued on 26 October 2005 and later amended on 17 November 2006. | 06.08.2009 |
| 10016739 | AEROMETALS | EASA.IM.R.S.01064 | | FAA STC SH3756NM | MD HELICOPTERS 369HE MD HELICOPTERS 369HS | Installation of Aerometals tailboom P/N 369X3500-505 or Yehner tailboom P/N YH-203. The present STC is the validation of the FAA STC SH3756NM amended on 30/11/99, limited to MDHI 369H, HS and HE models. | | Aerometals Tailboom P/N 369X3500-505- Installation Instructions: Aerometals Master Drawing List (MDL) No 25, revision B dated 22/10/99 approved by FAA or later FAA-approved revision. - Instructions for Continued Airworthiness: Aerometals Instruction (...) | 1. Service life of the Aerometals and Yehner referenced tailbooms is established according to FAR 27.571 (e) at 10000 flight hours, with periodic inspections required at 2450 flight hours and each 2450 flight hours thereafter. 2. Inspections of the (...) | 30.05.2005 |
| 10016740 | AEROMETALS | EASA.IM.R.S.01065 | | FAA STC SH5055NM | MD HELICOPTERS 369D MD HELICOPTERS 369E MD HELICOPTERS 369F MD HELICOPTERS 369FF | Installation of Aerometals tailboom: P/N 369X23500-505 as replacement of MDHI tailboom P/N 369D2300-507, or P/N 369X23500-507 as replacement of MDHI tailboom P/N 369D2300-507, or Yehner tailboom : P/N YH-228-1 as replacement of MDHI tailboom P/ (...) | | Installation Instructions and Instructions for Continued Airworthiness: see table in the paragraph "Limitations and Conditions" hereafter. | 1. Service life of the Aerometals and Yehner referenced tailbooms is established according to FAR 27.571 (e) at 10300 flight hours, with periodic inspections required at 7300 flight hours and each 1000 flight hours hereafter. 2. Inspections of the Ae (...) | 17.05.2005 |
| 10016931 | AEROMETALS | EASA.IM.R.S.01317 | | FAA STC SR01524SE | MD HELICOPTERS 369A MD HELICOPTERS 369HE MD HELICOPTERS 369HM MD HELICOPTERS 369HS | Horizontal stabilizer, P/N AM369A3600-603 Validation of FAA STC SRSR01524SE reissued June 28, 2006 | | Master Drawing List MDL-AMC-008, revision initial release, dated August 6, 2004 or later FAA approved revisions. AMC installation instructions MIO3003, revision initial release, dated December 15, 2003 or later FAA approved revisions. | Limited to MDHI (Hughes) models 369HM, 369HS, 369HE all other limitation and conditions listed in FAA STC SRSR01524SE reissued June 28, 2006 apply | 31.05.2007 |
| 10017013 | AEROMETALS | EASA.IM.R.S.01430 | | FAA STC SR00981LA | MD HELICOPTERS 369D MD HELICOPTERS 369E MD HELICOPTERS 369FF MD HELICOPTERS 369HE MD HELICOPTERS 369HM MD HELICOPTERS 369HS MD HELICOPTERS 500N | Strut Assy - Landing Gear, p/n 369XH6001, EASA validation of FAA approved PMA part. | | Instructions for Continued Airworthiness 369XH6001 Rev 03/18/05 or later FAA approved revision. Master Drawing List No. 28 Rev G 07/19/05 or later FAA approved revision. | | 29.04.2008 |
| 10017014 | AEROMETALS | EASA.IM.R.S.01431 | | FAA STC SR00766LA | MD HELICOPTERS 369D MD HELICOPTERS 369E MD HELICOPTERS 369FF MD HELICOPTERS 369HE MD HELICOPTERS 369HM MD HELICOPTERS 369HS | Skid Tube Position Light Assys., p/n 369XH92041, EASA validation of FAA approved PMA part. | | Aerometals Instructions for Continued Airworthiness AMI-11 Rev A 02/16/99 or later FAA approved revision. Master Drawing List No. 18 Rev A 11/11/99 or later FAA approved revision. | | 29.04.2008 |
| 10017015 | AEROMETALS | EASA.IM.R.S.01432 | | FAA STC SR01062LA | MD HELICOPTERS 369A MD HELICOPTERS 369D MD HELICOPTERS 369E MD HELICOPTERS 369HE MD HELICOPTERS 369HM MD HELICOPTERS 369HS MD HELICOPTERS 500N | Exhaust Tailpipe Outlet hangar Assys., p/n 369X8236, EASA validation of FAA approved PMA part. | | Aerometals Instructions for Continued Airworthiness AMI-18 Rev C 08/11/03 or later FAA approved revision. Master Drawing List No. 30 03/26/02 or later FAA approved revision. | | 29.04.2008 |
| 10017016 | AEROMETALS | EASA.IM.R.S.01433 | | FAA STC ST01550SE | MD HELICOPTERS 369D MD HELICOPTERS 369E MD HELICOPTERS 369FF MD HELICOPTERS 500N | Scavenge Air Elbow Assy., p/n AM369D290135, EASA validation of FAA approved PMA part. | | Installation Instructions AMC MI-04003 Rev IR 10/06/04 or later FAA approved revision. Master Drawing List MDL-AMC-019 Rev IR 10/06/04 or later FAA approved revision. | | 29.04.2008 |
| 10017019 | AEROMETALS | EASA.IM.R.S.01437 | | FAA STC SR01136SE | MD HELICOPTERS 369D MD HELICOPTERS 369E MD HELICOPTERS 369HM MD HELICOPTERS 369HS MD HELICOPTERS 500N | Landing Gear Filler Fairings p/n AM369A6200-45, -46, -57, -58; and Landing Gear Fairing Guides p/n AM369H6200-73, -74. EASA validation of FAA approved PMA part. | | MDHI AMM CSP-HMI-2 Rev 21 or later FAA approved revision. Master Drawing List AMC-003 Rev A 08/29/02 or later FAA approved revision. MDHI Service & Maintenance Instructions: - CSP-HMI-2 Section 32-10-00 (Models 369D, E, F & 500N) - CSP-H2 Section 6 (Models 369A, HM, HS, HE) | | 29.04.2008 |
| 10016708 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.00548 | | FAA STC SR02381AT | BELL 427 | Manufacture and Installation of Passenger Sliding Door for Bell 427 Model Helicopters (FAA STC SR02381AT dated 19/09/2002). | | -AAI Master Drawing List Report AA-98094 dated 14/04/1999. -Installation Instructions AA-98085 Revision B dated 29/11/2001. -Rotorcraft Flight Manual Supplement AA-99080 dated 2/10/2001. -Instructions for Continued Airworthiness AA-99043 dated 5/12/2001. | 1. Rotorcraft Flight Manual Supplement dated 7/09/2001, is a required part of this STC. 2. "Instruction for Continued Airworthiness", AAI Report No. AA-99043, Revision A dated 6/10/2000 is also a required part of this STC. 3. This STC is a (...) | 26.01.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|----------------------|---|--|---------------------|---|---|---------------|
| 10016762 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01091 | | FAA STC SR09324RC-D | BELL 407 | Installation of Passenger photo Window for Bell 407 Model Helicopters in accordance with:-the FAA STC SR09324RC-D dated 22/9/2003-AAI Master Drawing List Report AA-01095, Revision B, dated 19/3/2003. | | Instructions for continued airworthiness AA-01093 revision B dated 5/12/2001. Installation instructions Passenger Window Kit AA-01091 dated 19/9/2003, Drawing list AA-1095 dated 3/7/2001. | 1. The compatibility of this design change with other previously approved modifications must be determined by the installer.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the par (...) | 22.08.2005 |
| 10016767 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01096 | | FAA STC SR01450LA | BELL 407 | Installation of Apical Industries, Inc. Emergency Float Assembly, Part number 604.9501 for Bell 407 Model Helicopters in accordance with:-the FAA STC SR01450LA dated 27/5/2003.-Master document List MD407-11 Revision C, dated 28/4/2003. | | Apical Industries, Inc. Rotorcraft Flight Manual Supplement FMS 407 (11) FAA approved on 13/5/2003. Instructions for continued airworthiness ICA407-11 Revision A dated 19/5/2003. Approved landing gears for Apical Emergency Floats kit. Report No AA407-18 (...) | 1. The compatibility of this design change with other previously approved modifications must be determined by the installer.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the par (...) | 22.08.2005 |
| 10016770 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01102 | | FAA STC SR09239RC | BELL 407 | Installation of a Steerable Landing Light. | | FOO-12001 Revision B dated 15 October 1999 or later approved revision - Premier Aviation Master Drawing List.F09-62003 Revision - dated 16 March 2000 or later approved revision - Premier Aviation FMS.F08-72000 Revision - dated 2 April 1998or later accepted revision - Premier Aviation ICA. | None | 10.05.2007 |
| 10016771 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01105 | | FAA STC SR01849AT | BELL 212, 412, 412 EP | Light weight nitrogen reservoir kit/Replacement of nitrogen cylinder for emergency flotation system kit. | | AA-98143 Rev F, dated. Master Drawing list report or later approved revision AA-98142 installation instruction report and time limited parts. | | 20.12.2007 |
| 10016777 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01116 | | FAA STC SR02074AT | BELL 427 | Manufacture of Automatic Door Opener for Bell 427 Model Helicopters in accordance with the FAA STC SR02074AT. AAI Master Drawing List Report AA-98109 dated 30/10/1998. FAA STC SR 02074AT dated 14/02/2000 (Atlanta Aircraft Certification Office). | | - PMA No. PQ1058CE dated 15/02/2000.- Instructions for continued airworthiness AA-99043 dated 05/12/2001- Crew door opener kit AA-98099 dated 23/04/1999- Passenger door opener kit AA-98100 dated 12/12/2002- Baggage door opener kit AA-98105 dated 24/06/2002 | 1. The compatibility of this design change with other previously approved modifications must be determined by the installer.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph (...) | 15.07.2005 |
| 10016791 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01139 | | FAA STC SR01411AI | BELL 407 | Manufacture and installation of console assembly | | In accordance with Aeronautical Accessories Inc Drawing List Report No AA-97037, Rev A, dated April 23, 1997, and installation report No AA-96181, Rev A, dated March 6, 1997, or later approved revision. | | 26.01.2006 |
| 10016840 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01211 | | FAA STC SR01844AT | A109, A109A, A109A II, A109C A109E, A109K2, A119 | Installation of Tail Rotor Lockout Kit for Agusta A109, A109A, A109AII, A109C, A109E, A109K2, A119 Model Helicopter | | - Compliance Checklist, Agusta A109/119 Tail Rotor Pedal Lockout Kit- Report No. AA-93088, Rev. N, 2-21-02, "Drawing List for Tail Rotor Pedal Lockout Kit on Bell 206A, B, L, L-1, L-3, L-4, 407, OH-58, 427 Model helicopters, Agusta A109A, A109AII, A1 (...) | Compatibility of this design change with previously approved modifications must be determined by the installer. If the holder agrees to permit another person to use this certificatedo alter the product, the holder shall give the other person written evidence of that permission | 06.04.2006 |
| 10016860 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01236 | | FAA STC SR00235AT | 206A, 206B, 206L, 206L-1, 206L-2, 206L-3, 206L-4, 407 | Installation of Headset Retention Brackets | | Report No. AA-93041, Rev C, 4-18-06, Drawing List Headset Bracket Assembly for Bell Helicopter Textron Model 206A, B, L, L-1, L-3, L-4, 407 Model Helicopters. Report No AA-93042, Rev C, 10-02-00, Installation Instructions for Headset Bracket Assembly (...) | None | 29.08.2006 |
| 10016872 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01248 | | FAA STC SR02070AT | BELL 427 | Installation of Baggage Floor Protector | | Report No AA-98097, Rev E, 5-31-06, Drawing List Baggage Floor Protector on Bell Helicopter Textron Canada Limited 427 Model Helicopters. Report No AA-98091, Rev D, 5-5-06, Installation Instructions for Baggage Floor Protector kit and Supplemental Ba (...) | | 25.10.2006 |
| 10016902 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01279 | | FAA STC SR01687AT | BELL 407 | Manufacture of crew and passenger emergency push out windows | | In accordance with Aeronautical Accessories, Inc. Drawing List Report No. AA-97167, no revision, dated February 4, 1998, and installation per Report No. AA-97142, no revision, dated January 23, 1998, or later approved revisions. | | 16.11.2006 |
| 10016920 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01305 | | FAA STC SR09373RC-D | BELL 412, 412EP | Bell Helicopter Vibration monitor (BHVM) System | | AA-04118 original issue dated 22 June 2005 or later approved revision - Drawing List, AA-04049, Revision A, dated 14 June, 2005, or later approved revisions - Instructions for Continued Airworthiness. AA-04118 Original issue dated 22 June (...) | | 31.05.2007 |
| 10016921 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01306 | | FAA STC SR09374RC-D | 212, 412, 412 EP | Helicopter Emergency Egress Lighting (HEEL) System. This STC is classified as "Basic" by the FAA and is approved under the Post Type Validation Principles (PTVP). | | Aeronautical Accessories, Inc. Drawing List Report No. AA-02027, Revision A, dated September 5, 2002, or later FAA/DAS approved revision. Instructions for Continued Airworthiness provided in Aeronautical Accessories Inc. Report AA-02029, Revision C, (...) | | 25.04.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|----------------------|---------------------------|---|--|--|---|---------------|
| 10016922 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01307 | | FAA STC SR01278NY-D | 412 EP | L-3 Communications FA2300 Series Combined Cockpit Voice Recorder and Flight Data Recorder Kit. This STC is classified as #BASIC# by the FAA and is approved under the Post Type Validation Principles (PTVP). | | AA-05062 Revision H dated 16 June 2006 or later approved revision # Drawing ListAA-05153 Revision F dated 16 June 2006 or later approved revision # Installation ReportAA-05063 Original issue dated 27 July 2006 or later approved revision # Flight Man (...) | No credit is given for operational requirements that require both a CVR and FDR. | 04.05.2007 |
| 10016925 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01311 | | FAA STC SR01474NY | BELL 412 EP | Installation of Honeywell Mark XXII EGPWS for Bell Helicopter Textron 412EP helicopters. Validation of FAA STC SR01474NY dated January 9, 2007 | | AAI drawing list report 4123401-MDL01, revision F, dated 01-03-07 or later approved revisionAAI Rotorcraft Flight Manual Supplement document 4123401-FMS01, Rev C dated 06 June 2007, or later approved revisionsInstruction for continued airworthiness (...) | | 20.06.2007 |
| 10016930 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01316 | | FAA STC SR00790NY | BELL 407 | Installation of a SFIM Automatic Flight Control System (AFCS) on Bell 407 helicopters. EASA validation of FAA STC SR00760NY dated July 27, 2007. This STC is classified #Basic# by the FAA and is approved under the post type Validation Principles (PTVP) | | HAS Installation drawing list document MDL97-06-02, rev H, dated July 16, 2006 or later approved revisionHAS Installation instructions report No. IM-97-06-02, rev H, dated July 17, 2006 or later approved revisionAeronautical Accessories, Inc. Rotorc (...) | This approval is for VFR operations only. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. Installation of SFIM Force trim system in accordance with FAA STC SR00760NY (EASA.IM.RS.013 | 07.04.2008 |
| 10016967 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01368 | | FAA STC SR00778AT | BELL 206A/B/L/L-1/L-3/L-4 | Expanded Instrument Panel. This STC is classified as #BASIC# by the FAA and is approved under the Post Type Validation Principles (PTVP). | | AA-95023 # Drawing List, revision E, dated 23 February 2006 or later approved revision; AA-94004 # Installation Instructions, revision B, dated 1 March 2000 or later approved revision. | | 16.11.2007 |
| 10016969 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01370 | | FAA STC SH2616SO | BELL 206A/B/L/L-1/L-3/L-4 | Lower Console Assembly. This STC is classified as "BASIC" by the FAA and is approved under the Post Type Validation Principles (PTVP). | | AA-88094 - Drawing List, no revision, dated 8 August 1989 or later approved revision; AA-00132 - Instructions for Continued Airworthiness, no revision, dated 25 October 2000 or later approved revision. | | 09.10.2007 |
| 10016970 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01371 | | FAA STC SR1075AT | BELL 407 | Crew and Passenger Wedge Windows. | | AA-96053 - Drawing List, no revision, dated 16 May 1996 or later approved revision; AA-96054 - Installation Instructions, no revision, dated 16 May 1996 or later approved revision; AA-96055 - Instructions for continued Airworthiness, no revision, dated 16 April 1996 or later approved revision. | | 09.10.2007 |
| 10016985 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01390 | | FAA STC SR00760NY | BELL 407 | Installation of SFIM Force Trim System on Bell 407 helicopters. This STC is classified #Basic# by the FAA and is approved under the post type Validation Principles (PTVP) | | HAS Master drawing list document MDL97-05-01, rev G, dated October 12, 2006 or later approved revisionHAS Corporation Rotorcraft Flight Manual Supplement HAS-FMS-97-05, rev A, dated July 27, 2007 or later approved revisionInstructions for continued (...) | | 07.04.2008 |
| 10016986 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01391 | | FAA STC SR01285AT-D | BELL 407 | Rubber mounted replacement chin window. This STC is classified as "BASIC" by the FAA and is approved under the Post Type Validation Principles (PTVP). | | AA-96122 # Drawing List, revision H, dated 1 May 2007 or later approved revision; AA-96156 # Installation Instructions, revision J, dated 26 April 2007 or later approved revision; or AA 03112 # Installation Instructions, revision C, dated 1 May 2007 o (...) | | 04.12.2007 |
| 10017043 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01467 | | FAA STC SH8421SW | BELL 206B, 206L-3, 206L-4 | Steerable Nose Landing Light. This STC is classified as #BASIC# by the FAA and is approved under the Post Type Validation Principles (PTVP). | | 100-12005 revision N dated 11 February 2008 or later approved revision # Drawing List: AA-04025 no revision dated 13 February 2004 or later approved revision # Instructions for Continued Airworthiness | | 09.07.2008 |
| 10017047 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01471 | | FAA STC SH3316SO | BELL 212, 412, 412EP | Pushout Window Assembly | | AA-92063 revision F dated 11 May 2000 or later approved revision # Drawing List; AA-92057 revision K dated 7 January 2004 or later approved revision # Installation Instructions for Bell 412EP S/N 36191 and subs AA-98179 revision D dated 30 July 2002 or (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 18.06.2008 |
| 10017051 | AERONAUTICAL ACCESSORIES, INC | EASA.IM.R.S.01475 | | FAA STC SR09379RC-D | BELL 412 EP | Pilot heater failure monitor kit (412-081-001/002), Stand by attitude battery monitor kit (412-080-001), Crew flashlight holder kit (412-086-001). | | SR09379RC-D #Optional Accessories Kits# - RFM Supplements:# AA-07147, dated 14.04.2008 or later approved revision;# AA-08008, dated 14.04.2008 or later approved revision.- Instructions for Continued Airworthiness: Aeronautical Access (...) | 1. Applicable to Registration Marks I-O TEL.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect (...) | 05.03.2009 |
| 10027902 | AERONAUTICAL ACCESSORIES, INC | | REV. 1 | FAA STC SR01515AT | BELL 407 | Expanded Area Instrument Panel | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | AA-97129 revision F dated 28 May 2009 - Drawing List, Expanded Instrument Panel for Bell Helicopter Textron 407 Model Helicopters; AA97130 revision F dated 18 April 2008 - Installation Instructions for Expanded Instrument panel; AA-02052 revision A date (...) | None | 01.12.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|----------------------|---|---|--|--|---|---------------|
| 10027913 | AERONAUTICAL ACCESSORIES, INC | | REV. 1 | FAA STC SR09533RC-D | BELL 407 | Vertical Fin Cap with LED Light Kit | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AA-04133 revision A dated 18 February 2009 - Drawing List, Vertical FinCap with LED Light Kit; AA-97130 revision F dated 18 April 2008 - Installation Instructions for Expanded Instrument panel; AA-06111 no revision dated 23 February 2009 - Instructions (...) | None | 01.12.2009 |
| 10028641 | AERONAUTICAL ACCESSORIES, INC | | | | BELL 206 A & B | Installation of Aft Passenger Seat Shoulder Harness/Lap Belt Assembly for Bell Helicopter Textron 206 A / B Models | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AAI Report Number AA-88090, Revision H, dated 8-21-06, "Drawing List, Passenger Seat Shoulder Harness/Lap Belt Assembly for Bell Helicopter Textron 206 A / B Models" or later revisions of the above listed documents approved by EASA in accordance with EA (...) | None | 29.01.2010 |
| 10028796 | AERONAUTICAL ACCESSORIES, INC | | | FAA STC SR02162LA | BELL 407 | Installation of Mounting Provisions for a Tail Boom Radar Altimeter in accordance with Meeker Aviation Drawing List for Radar Altimeter Mount Kit, AA-TBA-16-010, and Document List of Radar Altimeter Mount Kit (>,<)-> AA-TBA-16-011. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Meeker Aviation Drawing List for Radar Altimeter Mount Kit, AA-TBA-16-010, Revision A, dated August 18, 2008- Document List of Radar Altimeter Mount Kit, AA-TBA-16-011, Revision R, dated February 20, 2008- Instructions for Continued Airworthiness Re (...) | This STC provides mounting provisions only; it does not allow the holder to install an altimeter or any wiring to power an altimeter. | 10.02.2010 |
| 10028797 | AERONAUTICAL ACCESSORIES, INC | | | FAA STC ST02121LA | BELL 407 | Installation of a Tail Boom Antenna Mount Kit in accordance with Meeker Aviation MDL for VHF Communication Antenna Mount Kit, AA-TBA-010. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Meeker Aviation MDL for VHF Communication Antenna Mount Kit, AA-TBA-010, Revision IR, dated January 15, 2008- Instructions for Continued Airworthiness Report Number AA-TBA-008, Rev IR, dated January 11, 2008, or later revisions of the above listed doc (...) | This STC provides mounting provisions only; it does not allow the holder to install an antenna or any wiring to power an antenna. | 10.02.2010 |
| 10028872 | AERONAUTICAL ACCESSORIES, INC | | | FAA STC SR09325RC | BELL 407 | Installation of Bulged Skylight in accordance with Aeronautical Accessories, Inc. Drawing List, Report No AA-01070 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Instructions for Continued Airworthiness Bulged Skylight Kit Report AA-01069, Rev B, dated 4-18-2007 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 16.02.2010 |
| 10028891 | AERONAUTICAL ACCESSORIES, INC | | | FAA STC SR09326RC | BELL 407 | Development and Installation of High Visibility Crew Door in accordance with Aeronautical Accessories, Inc. Drawing List, Report No AA-01073 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Drawing List, Report No AA-01073, Revision G, dated 10th, December 2009- Instructions for Continued Airworthiness, Report No AA-01072, Rev D, dated 7th, December 2009, or later revisions of the above listed documents approved by EASA in accordance with (...) | None | 17.02.2010 |
| 10028893 | AERONAUTICAL ACCESSORIES, INC | | | FAA STC SR09376RC-D | BELL 407 | High Visibility Door Kits in accordance with Aeronautical Accessories, Inc. Drawing List Report No AA-05155 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Drawing List Report No AA-05155, Revision K dated 14th, December 2009- Instructions for Continued Airworthiness for High Visibility Door Kit on Bell Helicopter Textron Model 407 Helicopters, Report No AA-5158, Rev B, dated 12th, October 2009, or lat (...) | None | 17.02.2010 |
| 10031747 | AERONAUTICAL ACCESSORIES, INC | | | FAA STC SR00646AT | AB412, AB412EP | Replacement Low Skid Landing Gear Cross-tubes | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AA-94042 revision L dated 2 March 2010 - Drawing List; AA-94043 revision L dated 1 May 2006 - Installation Instructions; AA-01136 revision F dated 22 April 2010 - Instructions for Continued Airworthiness or later revisions of the above listed documents a (...) | None | 09.09.2010 |
| 10033707 | AERONAUTICAL ACCESSORIES, INC | | | FAA STC SH1281SO | 206A, 206B, 206L, 206L-1, 206L-3, 206L-4, 407 | Maintenance Step. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | KR-154 revision S dated 10 November 2010 - Drawing List; AA-87036 revision R dated 4 April 2003 - Installation Instructions for Pre-Flight Step/Handle on Bell 206A and B models; AA-87040 revision C dated 4 October 1999 - Installation Instructions for Ma (...) | None. | 04.02.2011 |
| 10034699 | AERONAUTICAL ACCESSORIES, INC | | | FAA STC SR09536RC-D | 429 | Installation of Health and Usage Monitoring System (HUMS) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AA-09040 revision 0 dated 9 November 2010 - Flight Manual Supplement or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)-AA-09033 revision D dated (...) | The HUMS kit is installed on a non-required/non-hazardous basis. The utilization of the HUMS kit for maintenance credit is limited to: 1) main rotor track and balance; 2) tail rotor dynamic balance; and 3) fuselage vibration absorber tuning. HUMS data may (...) | 20.04.2011 |
| 10015732 | AERONAUTICAL ENGINEERS Inc. | EASA.IM.A.S.01754 | REV. 4 | FAA STC ST01827LA | B737-300 / B737-400 | BOEING 737-300/400 FREIGHTER WITH OPTIONAL CARGO CONTAINER CONFIGURATION. Modification of notch frames to accommodate 88x125" container indisposition 10 (rear) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CIRI H-01 Issue 1: Special Condition - EWIS ICA. The requirements for environmental protect (...) | AEI Cargo Modification Drawing List R-668, Revision L, dated June 21, 2010 for B737-300 aircraft - AEI Cargo Modification Drawing List R-811, Revision E, dated June 21, 2010 for B737-400 aircraft - AEI Inspection Instructions Document R-834, Revision (...) | Limited to B737-300/400 modified with EASA STC IM.A.S.01754 Rev 3 (FAAST01827LA) | 24.09.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|-------------------|----------------|--------------------------|---|---|--|---|---|---------------|
| 10015836 | AERONCA Inc. | EASA.IM.A.S.01928 | | ST09456SC | BOEING 737-300 BOEING 737-400 BOEING 737-500 | Manufacture and Installation of Engine Fan Cowl Doors | | Manufacture of Engine Fan Cowl Doors in accordance with Aerona Document No. TM518, "737 Cowl Door Drawing Control Document" Revision A dated February 19, 1998 or later approved revision/Installation of Engine Fan Cowl Doors in accordance with Aerona (...) | Compliance with Aronca Engineering Report Number ER 1904 "Instructions for Continued Airworthiness and Structural Repair Manual" Revision B, dated January 29, 1998, or later approved revision required. | 28.07.2006 |
| 10017069 | AEROQUEST INTERNATIONAL | EASA.IM.R.S.01505 | | TCCA STC SH97-63; | EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 EUROCOPTER FRANCE AS 350B EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 | AS350/AS355 Aeroquest Survey Installation | | IM-PA-06-040-91-01 issue 1 revision D dated 29 April 2008 or later approved revision # Installation manual;MDL-PA-06-040-91-01 issue 1 revision D dated 22 April 2008 or later approved revision # Master Drawing list;FMS-PA-06-040 issue 1 revision G d (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 28.05.2009 |
| 10017073 | AEROQUEST INTERNATIONAL | EASA.IM.R.S.01513 | | TCCA STC SH08-52 ISSUE 1 | EUROCOPTER FRANCE SA 315B | Installation of Aeroquest Survey Modification - Validation of Transport Canada (TCCA) STC No. SH08-52. | | JCM Aerodesign Limited Flight Manual Supplement 0630025FMS, dated 9-Oct.-2008 or later approved revisions;JCM Aerodesign Limited Maintenance Manual Supplement 0630025MMS, dated 9-Oct.-2008 or later approved revisions;JCM Aerodesign Limited Master Draw (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 06.02.2009 |
| 10014614 | Aeroservis s.r.o. | EASA.A.S.02741 | | | CESSNA T210 SERIES | Upgrade of NAV/COM equipment for IFR navigation. | | Pilot's operating handbook supplement No. 40, 41, 42, 43, 44, 45, 46, 47. | 1. S/N: 210638072. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 30.04.2007 |
| 10015076 | Aeroservis s.r.o. | EASA.A.S.03407 | | | L 200 A | Installation of NAV/COM/GPS device GNS 430 | | Supplement to AFM (only the Czech language) No. STC AE/079-DLP1 "Doplněk Smerice pro pilota letounu L200A". | | 25.03.2008 |
| 10027356 | AEROSERVIS, S.R.O. | P-EASA.A.S.03904 | | | CESSNA F172-N | Installation of NAV/COM Equipment:GNS 430 (NAV/COM/GPS), KX155 (NAV/COM), KR87 (ADF),KMA 24 (AUDIO/MKR) | FAR Part 23, Amdt. 50, dated 11 March 1996 | Pilot's Operating Handbook Supplements:- No: STC AE/088-DLP1 (GNS 430)- No: STC AE/088-DLP2 (KX 155)- No: STC AE/088-DLP3 (KR 87)- No: STC AE/088-DLP4 (KMA 24) | none | 24.09.2009 |
| 10027739 | AEROSERVIS, S.R.O. | P-EASA.A.S.03929 | | | L 410 UVP-E, L 410 UVP-E9 | The set of the safety modifications in order to prepare the alternative means of the compliance to EASA AD 2008-0104 and EASA AD 2008-0105. | FAR Part 23 Amdt. 34, dated 17 February 1987 through Amdt. 49, dated 11 March 1996. | L 410 UVP-E Flight Manual Supplement No. STC AE/090-DLP1L 410 UVP-E9 Flight Manual Supplement No. STC AE/090-DLP2L 410 UVP-E, E9 Wiring Manual Supplement No. STC AE/090-DAE1L 410 UVP-E, E9 Maintenance Manual Supplement No. STC AE/090-DTP1L 410 UVP-E, (...) | | 26.10.2009 |
| 10030622 | AEROSERVIS, S.R.O. | | | | L 410 UVP | The set of the safety modifications in order to prepare the alternative means of the compliance to the EASA AD 2008-0103. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval FAR Part 23 Amdt.34, dated 17 February 1987 through Amdt. 49, dated 11 March 1996 | L 410 UVP Flight Manual Supplement No. STC AE/103-DLP1, dated March 15/10-L 410 UVP Wiring Manual Supplement No. STC AE/103-DAE1, dated March 2010-L 410 UVP Maintenance Manual Supplement No. STC AE/103-DTP1, dated March 2010-L 410 UVP Maintenance Sche (...) | None | 28.06.2010 |
| 10030755 | AEROSERVIS, S.R.O. | P-EASA.A.S.03873 | | | SF340A | Installation of Universal UNS-1Msp Flight management System | | -, AFM Supplement STC AE/092-DLP1- ,Documentation in accordance with Master Documentation List AE 92 000X Rev. E, dated 2009-06.25 | Valid for Saab SF340A/s/n 340.026 and 340.038 only. | 03.07.2009 |
| 10015200 | AEROSPACE DESIGN AND CERTIFICATION | EASA.A.S.03592 | | | AIRBUS A320-232 | A320 In Flight Entertainment # IFE # Upgrade | | 1. Compliance Plan and Checklist, ref. ADC14/JN116, Issue3, dated 24th October 2008, 2. Master Data List, ref. ADC17/JN116, Issue 2, dated 23rd October 2009, 3. CAANZ Approved Flight Manual Supplement # Airbus A320-232 Panasonic EFX IFE Upgrade, re (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 29.10.2008 |
| 10015246 | AEROSPACE DESIGN AND CERTIFICATION | EASA.A.S.03683 | | | DIAMOND DA-42 | Banbury Modification 001 - FLIR Installation. | | JN147 ADC17 Issue 5 # Master Data List. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 07.11.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|------------------|----------------|----------------------|---------------------------------------|---|--|---|--|---------------|
| 10030093 | AEROSPACE DESIGN AND CERTIFICATION | | | | BAE 146-200 | New Cabin Configuration | JAR 25 Change 10The Certification Basis for the original product remains applicable to this certificate/ approval | Certification Program JN223ADC12(1) Issue 3or later revisions of the above listed documents approved by EASA | None | 21.05.2010 |
| 10031738 | AEROSPACE DESIGN AND CERTIFICATION | | | | B300 | Review of modifications installed prior to import | The Certification Basis for the original product remains applicable to this certificate/ approval | 1) Systems installation and testing in accordance with aerDAC FormJN222 ADC34 (1) Iss 1 dated 29/03/10) Operating Instructions defined in the following Airplane FlightManual Supplements:-a. Supplement No 001: aerDAC Form JN222 ADC29 (1)Iss 1 date (...) | Aircraft Serial Number FL-327 Only | 09.09.2010 |
| 10032496 | AEROSPACE DESIGN AND CERTIFICATION | P-EASA.R.S.01593 | | | AS 350 B, D, B1, B2, BA, BB AS 350 B3 | Installation of Indraero Siren IS10-100 Cargo Hook | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval | ADC12 issue 2 dated 8 October 2010 - Certification ProgramADC17 Issue 1 dated 8 October 2010 - Master Data ListADC54-27issue 1 dated 8 October 2010 - Instructions for ContinuedAirworthinessADC34A issue 1 dated 16 September 2010 - Installation Instructions | None | 11.11.2010 |
| 10032541 | AEROSPACE DESIGN AND CERTIFICATION | P-EASA.A.S.03947 | | | BOEING 767-341 | Review of modifications installed prior to import (B767-341Installationof SELL Galley, Crew Rest, 221 LOPA & ELT) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approvalCRI H-01 "Enhanced Airworthiness Programme for Aeroplane Systems - ICAon EWIS".The requ (...) | Master Data List, doc. JN189 ADC17 Issue 4 14/10/2010.Flight Manual Supplement, doc. JN189 ADC29 Issue 1 dated 16/11/09Instructions for Continued Airworthiness, doc. JN189 ADC54-25 Issue 5dated 14/10/2010Declaration of Compliance, doc. JN189 ADC58 I (...) | Limitations as per associated technical documentation:- Flight Manual Supplement, doc. JN189 ADC29 Issue 1 dated 16/11/09-Instructions for Continued Airworthiness, doc. JN189 ADC54-25 Issue 5dated 14/10/2010.Restricted to S/N 24753 | 15.11.2010 |
| 10034447 | AEROSPACE DESIGN AND CERTIFICATION | | REV. 1 | | B737-700 | Cabin certification for aircraft use in Commercial operations | The Certification Basis for the original product is unchanged and remainapplicable to this certificateThe requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate | Master Data List JN314-002 ADC17 Issue 2 dated 4/04/2011-AFM supplement JN314-002 ADC29 Issue 1 | Refer to AFM supplement JN314-002 ADC29 Issue 1 | 12.04.2011 |
| 10037643 | AEROSPACE DESIGN AND CERTIFICATION | | | | A330-243 | Installation of new Business Class Avio EBIZ Seats and IMS Rave IFESystem.The FWD cabin is currently configured with BCand EY seats which are tobe removed and replaced with new Avio BC seats. The MAS 200E IFE systemwill be removed from the FWD Cabin (...) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approval.A330 EASA TCDS A.004 and the following CRIs:-,F-4302-003 IFE ISPSS-,F-4301-003 Execu (...) | Certification Programme (including Master Data List in chapter 9)JN284-001 ADC12 (1) Issue 3 dated 1st of November 2011;or later revisions of the above listed documents approved by EASA. | None. | 13.12.2011 |
| 10038646 | AEROSPACE DESIGN AND CERTIFICATION | | | | MBB-BK 117 B-2 | Installation of LTS101-850B-2 Engines into MBB-BK117 B-2 Helicopter | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Master Data List JN216 ADC17, Issue 2, Rev 2 dated 29/02/2012- ICA JN216 ADC54-29 (1), Issue 1, Rev 3 dated 05/12/2011- RFMS JN216 ADC 29 (1) Issue 1, Rev 6, dated 01/03/2012- Installation of LTS101-850B-2 Engines into MBB-BK117B-2- RFMSJN216 ADC (...) | None | 05.03.2012 |
| 10039615 | AEROSPACE DESIGN AND CERTIFICATION | | | | B200 KING AIR | Installation of polyurethane reticulated foam into fuel tanks, asdefined by:JN323 ADC17 (1) Master Data List | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noiseand/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with JN323 ADC34 (1) Issue 1<->->Modification Instructions<->-> in accordance with JN323 ADC29 (1) Issue 1, Flight ManualSupplementMaintain in accordance with JN323 ADC54-23 (1) Issue 1<->->Instructions for Continu (...) | This STC grants no credit for any fuel tank flammability reduction. | 14.05.2012 |
| 10040189 | AEROSPACE DESIGN AND CERTIFICATION | | | | B757-200 | Installation of VIP Bedroom and Bathroom | EASA IM A.205 at Issue 2.The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approval.CRI D-01 SC for Isolated CompartmentsNon-significantThe requi (...) | AerDAC Master Data List No. JN312 ADC17(1), Issue 2, dated14 June2012.Instructions for Continued Airworthiness Including EWIS ICA No. JN312ADC54-25(1) Issue 1or later revisions of the above listed documents approved by EASA | None | 15.06.2012 |
| 10042466 | AEROSPACE DESIGN AND CERTIFICATION | | | | A340-300 | Installation of new Business Class Avio EBIZ Seats and IMS Rave IFESystem.The FWD cabin is currently configured with BCand EY seats which will beremoved and replaced with new Avio BC seats. The Rockwell Collins IFESystem will be removed from the FWD (...) | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | JN284-002 ADC17 (1) Issue 3 Master Data List dated 30th of November2012.JN284-002 ADC57 (1) Issue 1 Compliance Evaluation Report dated 4th ofApril 2012 or later revisions of the above listed documents approved by EASA | None | 03.12.2012 |
| 10043136 | AEROSPACE DESIGN AND CERTIFICATION | EASA.A.S.036S7 | | | DC9-32 | 110Y LOPA (Removal of one row of seats to improve access totype II exits to meet localretroactive requirements). | | Master Data List MDL/JN139, issue 1, dated 14 August 2008. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The inst (...) | 20.08.2008 |
| 10043534 | AEROSPACE DESIGN AND CERTIFICATION | | | | BELL 429 | Bell 429 Human External Cargo HookaeroDAC Modification DualCargo Hook Described in Master Data List JN382Issue 1, 10 January 2013 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI F-01HIRF ProtectionThe requirements for environmental protection and the associatedcertificated noise and/ or emissions (...) | Flight Manual Supplement No.1 aerDAC Modification NumberJN382 DualCargo Hook, ISSUE 1, dated 04 February 2013or later revisions of the above listed documents approved by EASA- Master Data List Dual Cargo Hook JN382 Issue 1, 10 January 2013. Inst (...) | -Hover performance data and procedures for dual cargo hooks operations are not provided.- Operation with persons attached to the dual cargo hook system shall not be conducted without approval by the responsible authority in accordance with the operati (...) | 04.02.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|----------------|----------------|----------------------|--|--|--|---|--|---------------|
| 10044760 | AEROSPACE DESIGN AND CERTIFICATION | | | | A330-243 | Installation of new Avio E-Biz BC Seats, Weber EY Seats, and IMS RavelFE System | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List, ref : JN424 ADC17 Issue 1 Instruction for Continued Airworthiness, ref : JN424 ADC54-25 Issue 1 Weight and Balance Report, ref : JN424 ADC20 Issue 1 for later revisions of the above listed documents approved by EASA, | None | 03.05.2013 |
| 10014369 | AEROSPACE DESIGN AND ENGINEERING | EASA.A.S.01640 | | | BOEING 737-3M8 | Introduction of G2 Stowage Galley | | Aerospace Design & engineering Consultants Ltd. Modification MOD-0016 | -- | 06.04.2006 |
| 10014371 | AEROSPACE DESIGN AND ENGINEERING | EASA.A.S.01645 | REV. 1 | | 737-76N/-600/-800/-900/-900ER | Installation of an Enhanced Surveillance Mode S Transponder | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amendment 12CS 25.869CS 25.1301CS 25.1307CS 25.1309CS 25.1321CS 25.1333CS 25.1351CS 25.1353CS 25.1355CS 25.1357CS 2 (...) | Aerospace Design & Engineering Consultants Modification MOD-0014 Aerospace Design & Engineering Consultants AFM Supplement 1 FMA-0014-001 Issue A Modification Summary MOD-1370 Revision A Flight Manual Supplement FMA-1370-083 Revision 1 for later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 29.07.2013 |
| 10014397 | AEROSPACE DESIGN AND ENGINEERING | EASA.A.S.01691 | | | BOEING 737 | Introduction of Three Overhead Liferaft Stowages | | Aerospace Design & Engineering Consultants Ltd Modification Summary MOD-0023 | none | 06.04.2006 |
| 10014405 | AEROSPACE DESIGN AND ENGINEERING | EASA.A.S.01700 | | | BOEING 737-200 BOEING 737-300 BOEING 737-400 BOEING 737-500 BOEING 737-600 | Installation of a Dual High Frequency (HF) Communication System. | | MOD-0001 Revision B, Modification Summary, dated 16 May 2007. | As referred in Paragraph 4.1 of MOD-0001 Revision B, Compatibility of this design change with other previously approved modifications must be determined by the installer. | 31.05.2007 |
| 10014459 | AEROSPACE DESIGN AND ENGINEERING | EASA.A.S.02517 | | | BOEING 737-4Y0 | Introduction of 8/150 Passenger Configuration and Associated Customisation | | - Modification Summary: MOD-0026 Issue C dated 7th September 2006, or later approved revision- Modification Certification Plan: MCP-0026 Issue C dated 7th September 2006, or later approved revision. | Compatibility of this design change with other previously approved modifications must be determined by the installer. | 20.10.2006 |
| 10014460 | AEROSPACE DESIGN AND ENGINEERING | EASA.A.S.02518 | REV. 1 | | BOEING 737-300 BOEING 737-400 BOEING 737-500 | Introduction of Enhanced Surveillance Mode S ATC System / Installation of new Mode S Transponders, components and data link wiring which will provide the ATC System with additional parameters necessary to comply with TGL N 18 revision 1, dated August 3 2003. | | - Flight Manual Supplement Ref: FMA-0031-003 Issue 1, or later approved revision- Modification Summary: MOD-0031 Issue A dated 21st February 2006, or later approved revision- Modification Certification Plan: MCP-0031 Issue A dated 21st February 2006, or later approved revision | 1. Compatibility of this design change with other previously approved modifications must be determined by the installer. | 07.09.2006 |
| 10014855 | AEROSPACE DESIGN AND ENGINEERING | EASA.A.S.03079 | | | BOEING 767-300 | Introduction of 25XF/188Y passenger configuration and associated customisation. | | Modification summary (MOD-0187, Revision B, 26/03/2007) | Serial Number 24952 only. | 14.06.2007 |
| 10014882 | AEROSPACE DESIGN AND ENGINEERING | EASA.A.S.03110 | | | BOEING B737-300 | Introduction of 138 Passenger Configuration and Associated Customizations | | - Modification Summary MOD-0217 Rev A, dated 02 July 2007, or later EASA approved revision.- Aircraft Flight Manual in accordance with paragraph 3.4 of Modification Summary MOD-0217 Rev A, dated 02 July 2007, or later EASA approved revision.- Weight & (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 20.09.2007 |
| 10014946 | AEROSPACE DESIGN AND ENGINEERING | EASA.A.S.03206 | REV. 1 | | BOEING 747-300 | Installation of an Elementary Surveillance Mode-S System in a Boeing 747-30. | | FMA-0234-031 Rev. 1 Flight Manual Supplement MCN 23410FMA 0234-032 Rev. 1 Flight Manual Supplement MCN 23751 | FMA-234-031 Rev. 1 Flight Manual Supplement MCN 23410FMA 0234-032 Ref. Flight Manual Supplement MCN 23751 This STC is approved only for the product configuration as defined in the approved design data. Compatibility with other aircraft/engine configuration shall be determined by the installer. | 02.11.2007 |
| 10014948 | AEROSPACE DESIGN AND ENGINEERING | EASA.A.S.03209 | | | DC-9-81 (MD-81) DC-9-81 (MD-82) DC-9-81 (MD-83) DC-9-81 (MD-87) | Elementary/ Enhanced Surveillance Mode S ATC System. | | Modification Summary MEO-0235 Rev. A, FMS No. 1 FMA-0235-033 Rev. 1, dated 25-07-2007 FMS No. 1 FMA-0235-034 Rev. 1, dated 25-07-2007 FMS No. 1 FMA-0235-036 Rev. 1, dated 25-07-2007 FMS No. 1 FMA-0235-05 Rev. 1, dated 25-07-2007 | | 11.01.2008 |
| 10015157 | AEROSPACE DESIGN AND ENGINEERING | EASA.A.S.03524 | | | BOEING 737-700 | Introduction of Seat-track mounted 40 Inch Closet-Boeing 737-700. | | Aerospace Design & Engineering Consultants Ltd. Modification Summary Document MCD-0352 Rev. A Aircraft Manual Supplement # MSN28630 Aircraft Manual Supplement # MSN28635 Component Maintenance Manual (CMM) 25-30-05 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 28.11.2008 |
| 10015242 | AEROSPACE DESIGN AND ENGINEERING | EASA.A.S.03676 | | | BOEING 767-200 | Introduction of 12C/241Y Passenger Cabin Configuration and Associated Floor Structure Changes. | | AeroDEC Modification Summary MOD-0477, Revision B | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 11.12.2008 |
| 10015279 | AEROSPACE DESIGN AND ENGINEERING | EASA.A.S.03804 | | | AIRBUS A320-200 | Introduction of 16J/126Y Class Cabin Configuration | | # Modification Summary, ref. doc. MOD-0548, Rev. A, dated 8th May 2009# Modification Compliance List, ref. doc. MCL-0548 Rev. A, dated 11th May 2009 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 11.05.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|------------------|----------------|----------------------|--|--|---|---|---|---------------|
| 10015297 | AEROSPACE DESIGN AND ENGINEERING | EASA.A.S.03847 | REV. 2 | | A318-111/-112/-121/-122 A319-115/-131/-132/-133 A319-111/-112/-113/-114 A320-111/-211/-212/-214 A320-215/-216/-231/-232/-233 A321-111/-112/-131/-211 A321-212/-213/-231/-232 | Installation of an Enhanced Surveillance Mode S System | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amendment 12 CS 25.669 CS 25.1301 CS 25.1307 CS 25.1309 CS 25.1321 CS 25.1333 CS 25.1351 CS 25.1353 CS 25.1355 CS 25.1357 CS 2 (-) | -Modification Summary, ref. doc. MOD-0544, Rev. C, dated 7th June 2010-Modification Certification Plan ref. doc. MCP-0544 Rev. B, dated 28th April 2009-Modification Compliance List, ref. doc. MCL-0544 Rev. C, dated 7th June 2010-Flight Manual Suppl. (...) | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product. | 29.07.2013 |
| 10027558 | AEROSPACE DESIGN AND ENGINEERING | P-EASA.A.S.03680 | | | DHC-8-102 | Traffic Alert and Collision Avoidance System (TCAS II) Installation. | The Certification Basis for the original product remains applicable to this certificate, CS25 elected to comply. | Aerospace Design and Engineering Consultants (AeroDEC®) Ltd. Modification Summary, MOD-0468, Rev. C or later approved Revision and Aerospace Design and Engineering Consultants (AeroDEC®) Ltd. Flight Manual Supplement No.1, FMA -0468-058 Rev. 2 or later (...) | | 09.10.2009 |
| 10027628 | AEROSPACE DESIGN AND ENGINEERING | P-EASA.A.S.03677 | REV. 1 | | A330-223 | Introduction of a change from a 48C/182Y dual class seating configuration to a 24C/242Y passenger seating configuration. | Certification Basis in accordance with the EASA Type Certification data Sheet TCDS A.004 issue 16 with CS25 Amendment 4 for the areas affected by the change (plus reversion to aircraft TC basis for §25.785 (3) (c)(2) head impact criteria). | -Modification Summary, doc. No MOD-0475 Revision D dated 15/10/09-Aircraft Manual Supplement Document No: AMS-330-294-299 dated 01/07/09 or later revisions of the above listed documents approved by EASA | None | 03.12.2009 |
| 10029548 | AEROSPACE DESIGN AND ENGINEERING | | REV. 1 | | B737-600, -700, -800, -900 | Change to STC "Introduction of a Dual Battery System, MOD-0702" The change consists in removing the EWIS limitation. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CRI H-01 Issue 2 /Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane (...) | -Modification Summary MOD-0702 Rev C dated 22 March 2010-EZA-0702 Rev A dated 10 November 2011 or later revisions of the above listed documents approved by EASA | The STC holder shall monitor all documentary changes of the basic AFM and related operational documentation issued by the TC holder to assure that these documents remain suitable to be used operationally for the installation of dual batteries under this STC. | 03.11.2011 |
| 10030206 | AEROSPACE DESIGN AND ENGINEERING | P-EASA.A.S.03549 | | | B747-400 | Reconfiguration of compartment zone A in accordance with Aerodoc Modification summary MOD-0396 Rev. B dated 29 March 2010 or later approved revision. This STC validates FAA STC ST01829SE. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CRI D-1 Width of Aisle for EASA.A.S.03549 | Jamco America Inc. Instructions for Continued Airworthiness, documents number JD-09AC03 Rev. New, dated March 06, 2008 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 02.06.2010 |
| 10032318 | AEROSPACE DESIGN AND ENGINEERING | | | | DHC-8-401, DHC-8-402 | Installation of Ground Recognition Lights | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Elected to comply with CS-25, Amendment 8 for changed areas. The requirements for enviro (...) | Aerospace Design and Engineering Consultants (AeroDEC®) Ltd. Modification Summary, MOD-0753, Rev. B Aerospace Design and Engineering Consultants (AeroDEC®) Ltd. Aircraft Manual Supplement No. AMS-DHC-8-Q 400-FLYBE, dated 01 August 2010 or later revisions of the above listed documents approved by EASA | none | 27.10.2010 |
| 10033237 | AEROSPACE DESIGN AND ENGINEERING | | | | B767-300 | Installation of First Class Seating | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this approval | Modification Summary, MOD-836, Revision A, 12/10/2010 or later revisions of the above listed documents approved by EASA | None | 05.01.2011 |
| 10039894 | AEROSPACE DESIGN AND ENGINEERING | | REV. 1 | | A330-243 | Revise the passenger seat place cabin configuration from 8FC/24JCL/183YCL (215 pax) to 8FC/247YCL (255 pax) passenger places together with associated changes to the complete aircraft interior. Installation of 64 passenger places of newly manufactured S (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. A330 EASA TCDS A.004 The requirements for environmental protection and the associated de (...) | Modification Compliance List Issue A, dated 30th May 2012-Modification Summary MOD-1083 Issue A, dated 29th May 2012 or later revisions of the above listed documents approved by EASA | None | 31.05.2012 |
| 10042785 | AEROSPACE DESIGN AND ENGINEERING | | | | 747-400 | Certification of drainmast p/n 24E507589G02 against 25.1455 on Boeing 747-467ERF MSN 37299 in order to nullify Hong Kong Civil Aviation Department Variation No VAR/2012/133. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | MCP-1216 Rev B (Modification Certification Plan) MCR-1216 (Modification Classification Request) MOD-1216 Rev A (Modification Summary) MCL-1216 Rev B (Modification Compliance List) or later revisions of the above listed documents approved by EASA | STC is only valid for B747-467F MSN37299 | 18.12.2012 |
| 10044229 | AEROSPACE DESIGN AND ENGINEERING | | | | B767-200/-300/-300F/-400ER | Introduction of Controller-Pilot Data-Link Communications | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): CRI F-01 "Data Link Services for the Single European Sky CS 25 Amendment 12: 25.601(->,<-) 25.611, 25. (...) | Aerospace Design & Engineering Consultants Ltd Modification Summary MOD-1282 Revision A-Aerospace Design & Engineering Consultants Ltd Flight Manual Supplement FMA-1282-084 Revision 1 or later revisions of the above listed documents approved by EASA | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 27.03.2013 |
| 10017117 | AEROSPACE DESIGN FACILITIES Ltd. | EASA.R.S.01099 | | | EUROCOPTER FRANCE AS 355 SERIE | Installation of Wescam nose mount | | Aerospace Design Facilities Modification Summary Sheet ADF 2004-001 Issue 3 dated July 2 2005 or later approved revision. Aerospace Design Facilities AFM Supplement 7 & 8, Issue 2, Dated June 2005 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.08.2005 |
| 10017147 | AEROSPACE DESIGN FACILITIES Ltd. | EASA.R.S.01250 | | | EUROCOPTER FRANCE AS 350 SERIE | Installation of Wescam combined nose and side mount | | Aerospace Design Facilities Ltd Flight Manual Supplement 9 | Approval limited to nose mount only | 01.03.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|------------------|----------------|----------------------|---|---|---|---|---|---------------|
| 10017276 | AEROSPACE DESIGN FACILITIES Ltd. | EASA.R.S.01512 | | | EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 EUROCOPTER FRANCE AS 355N | ADF2003-001 Utility Pole Mount. This STC supersedes EASA approval 2004-1997, dated 5 March 2004. | | Flight Manual Supplements: ADF2003-001 FM4 (AS355N), ADF2003-001 FM5 (AS355Fseries) | None. | 21.08.2008 |
| 10028585 | AEROSPACE DESIGN FACILITIES Ltd. | P-EASA.R.S.01444 | REV. 2 | | 1. AS350 B, B1, B2, BA, BB, BB, B3 1. AS350 D, EC130B4 2. AS355E, 2. AS355F, AS355F1, AS355F2, 2. AS355N, AS355NP | Cowling - Secondary Latching | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Modification Definition Aerospace Design Change Instruction Sheet ADF2007-021 Issue 1 Flight Manual Supplements Not Applicable. Maintenance Aerospace Design Airworthiness Inspection Schedule ADF/AIS2007-021 issue3 | None | 14.09.2010 |
| 10013963 | AEROSPACE ENGINEERING SOLUTIONS | EASA.A.S.00652 | REV. 2 | | BOEING 767-300 | Introduction of Matsushita In-Flight Entertainment System | | Aerospace Engineering Solutions Master Data List AES-767-367MDL or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 02.11.2005 |
| 10014122 | AEROSPACE ENGINEERING SOLUTIONS | EASA.A.S.01232 | REV. 1 | | BOEING 767-300 | Introduction of 258 Passenger Configuration & Associated Cabin Changes, Rev. 1: FAR Part 26 Validation for EASA STC EASA.A.S.01232. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AES Ltd. Master Data List No. AES-767-330MDL Issue 6 dated 14 December 2012 or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 23.01.2013 |
| 10014123 | AEROSPACE ENGINEERING SOLUTIONS | EASA.A.S.01233 | | | BOEING 767-300 | Introduction of Matsushita In-Flight Entertainment System | | AES Master Data List AES-767-356MDL Issue 2 or later EASA approved revision | | 30.05.2005 |
| 10014557 | AEROSPACE ENGINEERING SOLUTIONS | EASA.A.S.02655 | | | BOEING 767-300 | Introduction of Main/APU Battery Paralleling for Extended Standby Power Time Capability. | | Aerospace Engineering Solutions Modification AES-767-469 | Restricted to Aircraft G-DBLA ASN 26063 & G-PJLO ASN 26064. | 10.10.2006 |
| 10014854 | AEROSPACE ENGINEERING SOLUTIONS | EASA.A.S.03078 | | | B767 | Introduction of 290 Pac Place Interior Configuration | | Modification AES-767-787 Reports: AES-TR-0250, AES-TR-0251, AES-TR-0252, AES-TR-0258, AES-TR-0277. | S/N 23765 ONLY. | 31.05.2007 |
| 10014949 | AEROSPACE ENGINEERING SOLUTIONS | EASA.A.S.03212 | | | BOEING 757-200 | Introduction of alternative EGPWS Mode 6 Callout Menu. | | AES Modification AES-757-894, issue 2 01/08/2007 AES AFM Supplement No.6 issue 1 | FAA STC 00381SE must be installed prior to the installation of this STC | 21.08.2007 |
| 10015236 | AEROSPACE ENGINEERING SOLUTIONS | EASA.A.S.03662 | | | BOEING 757-200 | Introduction of Flight Control Computer (FCC) Interlock2. | | AES Master Data List, AES-757-1187MDL, Issue 1, 04/08/2008. | # Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. (...) | 13.08.2008 |
| 10034052 | AEROSPACE ENGINEERING SOLUTIONS | P-EASA.A.S.03621 | REV. 1 | | B757-200 | Introduction of IFE System Circuit Breaker, Relays and Wiring. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. For changed areas the defined certification basis is CS25 Amendment 4 CRI H-01 ICA on for EWIS The (...) | AES Master Data List, AES-757-1072MDL, Issue 3, 16/09/2010 or later revisions of the above listed documents approved by EASA | None | 02.03.2011 |
| 10037704 | AEROSPACE ENGINEERING SOLUTIONS | | | | A319-132 | Introduction of Panasonic dMPES In Flight Entertainment - IFE System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 "Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS" The req (...) | Modification Summary, ref. doc.: AES-319-1849, Issue 2, dated 14th December 2011- Master Data List, ref. doc.: AES-319-1949MDL, Issue 2, dated 14th December 2011 or later revisions of the above listed documents approved by EASA | None | 16.12.2011 |
| 10040625 | AEROSPACE ENGINEERING SOLUTIONS | | | | 321-231 | Introduction of Panasonic MPES IFE system | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Project Specific Certification Plan, ref. doc. AES-TR-1024, Issue 2 dated 15/05/2012- Master Data List, ref. doc. AES-321-2443MDL, Issue 1, dated 29/06/2012- Instructions for Continued Airworthiness - Principal Document, ref. doc. AES-TR-1031, Issue (...) | The approval is limited to MSN 2730 and MSN 2793 | 16.07.2012 |
| 10043054 | AEROSPACE ENGINEERING SOLUTIONS | EASA.A.S.00612 | REV. 1 | | BOEING 767-300 | Installation of Enhanced Ground Proximity Warning system (EGPWS), Rev. 1: FAR Part 26 Validation for the EASA STC EASA.A.S.00612. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - AES-767-346 Issue 1 - AFM Supplement No 4 Issue 1; AES Ltd. Master Data List No. AES-767-346MDL Issue 5 dated 30 October 2012; or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 23.01.2013 |
| 10043722 | AEROSPACE ENGINEERING SOLUTIONS | | | | B767-300 | Introduction Of "RAVE" IFE System. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AES Ltd. Master Data List No. AES-767-2525 MDL Issue 4 dated 18 February 2013; or later revisions of the above listed documents approved by EASA. | The System Control Unit (SCU) Sim Card must be removed. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 19.02.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|------------------------|----------------|-------------------------------|--|---|--|---|---|---------------|
| 10044405 | AEROSPACE ENGINEERING SOLUTIONS | | | | B767-38E | Introduction of Crew Rest Bunks | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s); CRI D-01 Installation of Crew Rest Compartments CS-25 Amendment 11 - Various The requirements for env (...) | AES Ltd Master Data List No. AES-767-2349 MDL Issue 3 dated 5 April 2013 AES Instruction for Continued Airworthiness No. AES-TR-1062 Issue 1 for later revisions of the above listed documents approved by EASA | Restricted to ASN 30840 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airw (...) | 10.04.2013 |
| 10027883 | AEROSPACE ENGINEERING SOLUTIONS LIM | | REV. 1 | | A320-214 | Introduction of Rockwell Collins dPAVES IFE System Revision of the EASA STC 10027883 to remove EWIS limitation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-01 "Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS" The re (...) | Modification Summary ref. Modification No. AES-320-1522, Issue 4, dated 21st March 2011 - Master Data List, ref. doc. No. AES-320-1522 MDL, Issue 4, dated 23rd April 2010 - Project Specific Certification Plan (PSCP), ref. doc. No. AES-TR-0600, Issue 4, (...) | Not applicable | 22.03.2011 |
| 10031517 | AEROSPACE ENGINEERING SOLUTIONS LIM | | | | B757-220, MSN 27973 | Introduction of Driessen G1A and G4B Galleys | FAA TCDS No. A2NM The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are (...) | AES Ltd MDL No. AES-757-1771 MDL Issue 4 dated 15 August 2010 or later revisions of the above listed documents approved by EASA | Restricted to ASN 27973 | 20.08.2010 |
| 10036814 | AEROSPACE ENGINEERING SOLUTIONS LIM | CAA UK CAA.21NE2.00057 | REV. 3 | CAA UK CAA.21NE2.00057 REV. 2 | B757-204/-236 | FAR Part 26 Validation for EASA STC CAA.21NE2.00057 | FAA TCDS A2NM The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Generic CRI H-01 EWIS The requirements for environmental protection and th (...) | AES Ltd Master Data List No. AES-757-172MDL, Issue 1, dated 20 September 2011 or later revisions of the above listed documents approved by EASA | None | 07.10.2011 |
| 10040704 | AEROSPACE ENGINEERING SOLUTIONS LIM | | | | A321 | Introduction of Twin Class Cabin Interior Configuration. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval plus: 1., CRI D-3306-001 "Application of Heat Release and Smoke Density Requirements to (...) | Modification Summary ref. Modification No. AES-321-2059, Issue 2, dated 19th July 2012; Master Data List, ref. doc. No. AES-321-2059 MDL, Issue 2, dated 19th July 2012 - Project Specific Certification Plan (PSCP), ref. doc. No. AES-TR-0844, Issue 3, (...) | The EASA STC 10038683 is a prerequisite for the installation of this STC. | 20.07.2012 |
| 10014059 | AEROSPACE ENGINEERING SOLUTIONS Ltd | EASA.A.S.01143 | | | B757-200 | Introduction of Pegasus Flight Management Computers & MultiMode Receivers (MMRs) | | AES Master Data List AES-757-376 MDL Issue 1 or later approved revision | | 23.03.2005 |
| 10014208 | AEROSPACE ENGINEERING SOLUTIONS Ltd | EASA.A.S.01360 | | | B767-300 | Introduction of 279 passenger configuration | | Aerospace Engineering Solutions Modification AES-767-426 dated April 13 2005 or later approved revision | | 01.07.2005 |
| 10014495 | AEROSPACE ENGINEERING SOLUTIONS Ltd | EASA.A.S.02557 | | | B767-300 | Introduction of 290 passenger place interior configuration | | Aerospace Engineering Solutions Modification AES-767-492 | | 22.05.2006 |
| 10043151 | AEROSPACE ENGINEERING SOLUTIONS Ltd | EASA.A.S.C.01062 | | EASA.A.S.00251 | BOEING 757-200 | Aerospace Engineering Solutions Modification AES-757-165 Issue 1 Introduction of Enhanced Ground Proximity Warning System (EGPWS) | | Aerospace Engineering Solutions AES-757-165 Issue 1 (dated June 02 2005) or later EASA Approved Revision. AES AFM Supplement Number 3 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall (...) | 16.08.2005 |
| 10016714 | AEROSPACE FILTRATION SYSTEMS, Inc. | EASA.IM.R.S.01008 | REV. 1 | FAA STC SR09421RC | BELL 206L-3 BELL 206L-4 | Installation of Aerospace Filtration System (AFS) Engine Inlet Barrier Filter (IBF) system in accordance with the FAA STC SR09421 RC. AFS Drawing Number 10600, Revision B dated 17/6/2004 FAA STC SR09421RC dated 25/08/2004 (Rotorcraft Certification Office Southwest Region) | | Installation Procedure AFS-BH206L3L4-IBF-KIT-IP, initial release dated 16/8/2004 or later approved - Rotorcraft Flight Manual Supplement (RFMS) AFS-BH206L3L4-IBF-KIT-FMS initial release dated 25/8/2004 or later approved - Instructions for continued air (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 27.01.2005 |
| 10016715 | AEROSPACE FILTRATION SYSTEMS, Inc. | EASA.IM.R.S.01009 | | FAA STC SR09368RC | BELL 407 | Installation of Aerospace Filtration System (AFS) Engine Inlet Barrier Filter (IBF) system in accordance with the FAA STC SR09368RC. AFS Drawing Number 10600, Revision A dated 8/4/2003 FAA STC SR09421RC dated 25/08/2004 (Rotorcraft Certification Office Southwest Region) | | Installation Procedure AFS-BH407-IBF-KIT-IP, initial release dated 11/4/2003 - Rotorcraft Flight Manual Supplement (RFMS) AFS-BH407-IBF-KIT-FMS initial release dated 10/12/2003 - Instructions for continued airworthiness (ICA) AFS-BH407-IBF-KIT-ICA initial release dated 4/12/2003 | 1. Rotorcraft Flight Manual Supplement dated 10/12/2003 is a required part of this STC2. "Instructions for continued airworthiness" AFS-BH407-IBF-KIT-ICA is also required part of this STC.3. This STC is approved only for the product configuration as (...) | 27.01.2005 |
| 10016741 | AEROSPACE FILTRATION SYSTEMS, Inc. | EASA.IM.R.S.01067 | | FAA STC SR01394LA | MD HELICOPTERS 369D MD HELICOPTERS 369E MD HELICOPTERS 369F MD HELICOPTERS 369FF MD HELICOPTERS 369H MD HELICOPTERS 369HE MD HELICOPTERS 369HM MD HELICOPTERS 369HS | Installation of Aerospace Filtration Systems Inlet Barrier Filter System instead of Engine Air Separator (with or without Mist Eliminator) or FOD Screen. According to FAA STC SR01394LA. Master Data List: MD500-IBF-KIT-MDL, Rev. D, dated 30.09.2004 or (...) | | Master Data List: MD500-IBF-KIT-MDL, Rev. D, dated 30.09.2004 or later FAA approved revision. Flight Manual Supplement MD500-IBF-KIT-FMS Rev. B FAA approved 12 April 2005 or later FAA approved revision. Operation & Maintenance Manual/ IPL MD500-IBF-KI (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 16.08.2005 |
| 10016929 | AEROSPACE FILTRATION SYSTEMS, Inc. | EASA.IM.R.S.01315 | | FAA STC SR09451RC | BELL 206B | Installation of Aerospace Filtration Systems (AFS) Engine Inlet Barrier Filter (IBF) on Bell 206B helicopters. Validation of FAA STC SR09451RC dated December 14, 2005. This STC is classified as #BASIC# by the FAA and is approved under the post type Validation Principles (PTVP) | | AFS drawing number 109000, revision A, dated August 16, 2005 or later approved revision. Installation procedures AFS-BH06B-IBF-KIT-IP, revision A, dated October 25, 2005 or later approved revision. Instruction for continued airworthiness document AFS- (...) | | 05.12.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|----------------------|----------------|----------------------|--|--|---|---|---|---------------|
| 10017072 | AEROSPACE FILTRATION SYSTEMS, Inc. | EASA.I.M.R.S.01509 | | FAA STC SR02393CH | EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA | Installation of Inlet Barrier Filter system in accordance with AFS, Inc. Master Data ListThis STC is the validation of FAA STC SR02393CH (issued on March 20, 2007) | | Master Data List No. AFS-AS350B-IBF-MDL Revision F dated 5 May 2008 or later EASA approved revision, Rotorcraft Flight Manual Supplement No. AFS-AS350B-IBF-FMS Revision B dated 8 Sep. 2008 or later EASA approved revision, Instructions for Continued Air (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 13.01.2009 |
| 10026895 | AEROSPACE FILTRATION SYSTEMS, Inc. | P-EASA.I.M.R.S.01495 | | FAA STC SR02526CH | MD900 | Installation of engine inlet barrier filter (IBF) system | The Certification Basis for the original product remains applicable to this certificate/ approval | Master Data List Inlet Barrier Filter system AFS-MD900-IBF-MDL Rev D dated 15 July 2009 - Aircraft Flight Manual Supplement AFS-MD900-IBF-FMS Rev A approved by the FAA on 21 July 2009 - Instructions for Continued Airworthiness AFS-MD900-IBF-ICA Rev (...) | None | 20.08.2009 |
| 10036331 | AEROSPACE FILTRATION SYSTEMS, Inc. | P-EASA.I.M.R.S.01331 | | FAA STC SR02338CH | A119 & AW119MKII | EASA Validation of FAA STC SR02338CH "Installation of a foreign object damage (FOD) screen or an inlet barrier filter (IBF) system" | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Rotorcraft Flight Manual Supplement is the document AFS-AA119EASA-IBF-FMS Revision I or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)/Air Filtrati (...) | None | 01.09.2011 |
| 10034213 | AEROSPACE TECHNOLOGIES GROUP, INC. | | | FAA STC ST03910AT | A320-111, -211, -212, -214 A320-215, -216 A320-231, -232, -233 | Installation of Passenger Cabin electrically powered Window Shades | The Certification Basis for the original product remains applicable to this certificate/ approval | FAA STC ST03910AT- Aerospace Technologies Group (ATG) Engineering Drawing List, TS-EDL-48-001, Rev. G dtd. 2 Nov 2010- Aerospace Technologies Group Document PRD-CAW-48-001, Rev. E dtd. 28 Feb 2011 or later revisions of the above listed documents appr (...) | None | 16.03.2011 |
| 10015550 | AEROSYS Inc. | EASA.I.M.A.S.01444 | | FAA STC ST01843LA | BOEING MD 11F | Installation of the main deck cargo compartment one minute smoke detector in accordance with FAA STC ST01843LA or later EASA-approved revision. | | See item 5 of approval report (attachment) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 16.02.2006 |
| 10034628 | AEROTEC - T. UHL GMBH & CO. KG | P-EASA.A.S.03924 | REV. 2 | | CESSNA 206H, T206H | Sky Diving Kit for Cessna 206H Installation of components to release sky divers. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Pilot's Operating Handbook Supplement, Doc. No. PO-206H-530001-Installation and Continuing Airworthiness Instructions, Doc. No. MM-206H-530001 or later revisions of the above listed documents approved by EASA. | None | 27.08.2012 |
| 10013984 | AEROTEC CONCEPT | EASA.A.S.01016 | | | BEECH C90 A | Installation of Garmin GNS 430 (VHF NAV ILS GPS/BRNAV), Allied Signal ART 2000 (weather radar), Allied Signal KMD 850 (MFD), MD 41 (panel annunciator) and GMA 340 (markers) | | -Dossier de modification Aerotec No 4104 | 1. Modification limited to s/n LJ 10982. EASA Supplemental Airplane Flight Manual GNS 430 must be used with this modification. 3. EASA Supplemental Airplane Flight Manual KMD 850 must be used with this modification. | 19.01.2005 |
| 10014014 | AEROTEC CONCEPT | EASA.A.S.01057 | | | PIPER PA-28RT-201T | Installation of GARMIN GNS 430 (V/JF/NAV/ILS/GPS) and MD 41 (panel annunciator) Removal of VHF KY 97A, GPS 150XL, NAV VIR351 and Glide GLS 350. Installation of GNS 430 and MD 41 panel annunciator, auto-pilot coupled. | | AFMS "Supplément au manuel de vol du PA 28 RT 201 T immatriculé F-GJRR pour une utilisation GNS 430 en moyen supplémentaire de navigation IFR et B-RNAV" dated 05 September 2005 | The approval is limited to MSN 28R7931294 F-GJRR | 25.10.2005 |
| 10014061 | AEROTEC CONCEPT | EASA.A.S.01147 | | | PIPER PA-31T | Installation of radio navigation and communication systems in accordance with Document Reference 5304 - Revision 2 dated April 15, 2005 | | AFMS referenced 5304, STC Document Reference 5304 Revision 2 dated April 15, 2005 | This STC is limited to Piper PA-31T SN 8020037 | 11.10.2005 |
| 10014106 | AEROTEC CONCEPT | EASA.A.S.01211 | | | BEECH 95-B55 | Radio and navigation change with BRNAV capacity and STEC 55X automatic pilot steered | | Modification File No. AEROTEC STC0505 Airplane Flight Manual Supplement No Modification 0505, edition 1, revision 0, dated 31st March 2005 or later approved revision | Modification limited to s/n TC1143AFMS No modification 0505 must be on board | 26.09.2006 |
| 10014278 | AEROTEC CONCEPT | EASA.A.S.01514 | | | PIPER PA-31 | Installation of Radio Navigation and Communication Systems-KX 175B, KY 196, KN 53, KN 75, KMA 24 and KN 65 removal-GNS 430, KX 155A, MD 41, RDR 2000, GMA 340, KN 62A, KI 229, G106A installation | | Airplane Flight Manual Supplement: Supplément au manuel de vol Piper PA 31 pour l'utilisation du GPS GNS 430 certifié en moyen supplémentaire de navigation IFR/VFR avec la capacité Basic BRNAV, installé suivant la modification 3005' dated 10/11/2005. | The approval is limited to MSN 31-367 F-GDPD | 20.02.2006 |
| 10014301 | AEROTEC CONCEPT | EASA.A.S.01545 | | | BEECH C90 | Installation of radio communication systems Model Beechcraft C90 SN LJ 710 | | Supplemental AFM for GNS 530 edition 1 of 15 January 2005, revision 2, Supplemental AFM for GNS 430 edition 1 of 15 January 2005, revision 2 and Supplemental AFM for KMD 850 edition 1 of 15 January 2005, revision 2 | 1. This STC is limited to Beechcraft C90 SN LJ 710.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft configurations shall be determined by the installer. | 12.09.2005 |
| 10014391 | AEROTEC CONCEPT | EASA.A.S.01677 | | | MOONEY M20E | BRNAV capacity with AP Steering | | Modification No. 0205/Aérotec Airplane Flight Manual supplement BRNAV Garmin GNS 430 - édition 1 dated 12 october 2005 | Modification limited to s/n 28R7931294 The AFMS BRNAV Garmin 430 must be used with this modification | 04.05.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------|------------------|----------------|----------------------|--|---|--|---|--|---------------|
| 10014756 | AEROTEC CONCEPT | EASA.A.S.02944 | | | CESSNA 208 CESSNA 208A CESSNA 208B | Video Mast Installation under Cessna 208. | | Installation must be in accordance with Aerotec Master Document List 7606 MDL Rev. 02, dated 22-08-2007, or later approved revisions. Operations must be in accordance with Flight Manual Supplement Aerotec 7606 SAFM Rev. 01, dated 17-08-2007, or late (...) | | 28.08.2007 |
| 10014794 | AEROTEC CONCEPT | EASA.A.S.02996 | | | PIPER PA-31T | Installation ATC mode S. | | Master Data List ref. "STC groupe 2 n° 1407", revision 0, dated August 2, 2007 or later approved revision. | 1. This STC is limited to the S/N: 7720036 and 8020037.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce (...) | 10.09.2007 |
| 10014799 | AEROTEC CONCEPT | EASA.A.S.03002 | | | PIPER PA-31T | Modification radio installation. | | # Master Data List ref. "STC groupe 2 n° 1507", revision 0, dated August 2, 2007 or later approved revision. # Airplane Flight Manual Supplement, ref. 1507-AFMS, revision 0, dated January 25, 2007 or later approved revision. | 1. This STC is limited to the S/N: 7720036.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse e (...) | 10.09.2007 |
| 10014985 | AEROTEC CONCEPT | EASA.A.S.03277 | | | BRITTEN NORMAN BN-2T ISLANDER | Survey Capacity on BN-2T | | Modification file 7307 according Drawing List No. 7307DL, revision 00, dated 25 August 2007 Installation Manual No. 7606-IM, revision 00, dated 5 February 2008 Instructions for Continued Airworthiness, revision 00, dated 26 January 2008 Supplemental A (...) | Supplemental Airplane Flight Manual, revision 00, dated 17 January 2008, or later approved revision must be on board when modification is installed. Instruction for Continued Airworthiness, revision 00, dated 26 January 2008 and Supplemental Airplane (...) | 08.04.2008 |
| 10015093 | AEROTEC CONCEPT | EASA.A.S.03430 | REV. 1 | | C-212-DD | Navigation / Communication (NAV/COM) set replacement and HF Communication set (provisions) installation. This STC replaces a previous NAV/COM system by an up-to-date one, and installs a new HF communication set (provision). This revision restores the aircraft (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-25 Amndt. 4A.C.43.13 The requirements for environmental protection and the associated (...) | Master Data List, ref : AE 0803 MDL Rev 4, dated 04 April 2009 ("") except for AFM Supplement and ICA, which are approved separately. AFM Supplement, ref : AE 0803 SMV Rev 4, dated 19 November 2008 Instruction for Continued Airworthiness, ref : AE 0 (...) | 1. This STC is limited to the MSN 3872. HF communication set (subpart of this STC) is not approved (not installed / not operational # wiring provisions are accepted) | 01.06.2011 |
| 10017145 | AEROTEC CONCEPT | EASA.R.S.01248 | | | EUROCOPTER FRANCE EC 120 B | Modification des sources d'altitude, d'écart de cap et GPS du système pilote automatique PA85 SAGEM (change of sensors sources for altitudes retention, heading error select and GPS steering of Sagem PA85 Automatic Pilot system) | | -Definition and Installation: according to AEROTEC technical file ref. 1106 rev.1- Inspection and Maintenance: refer to EASA.IM.R.S.1056 Rev.1 and EASA.R.C.01912 and EASA.R.C.01913 changes whose maintenance directives remain unchanged | Limited to aircraft EC120B s/n1327, F-GJIR only, when equipped with EASA.IM.R.S.1056 Rev.1 and EASA.R.C.01912 and EASA.R.C.01913 changes | 19.01.2007 |
| 10017169 | AEROTEC CONCEPT | EASA.R.S.01277 | | | EUROCOPTER FRANCE EC 120 B | Improvement of "VG Fail" alarm for EASA.R.S.01017 and EASA.IM.R.S.01056 installations | | Definition and Installation: AEROTEC technical file ref. FAM2906 Inspection and Maintenance: according to STCs EASA.R.S.01017 and EASA.IM.R.S.01056 directives that remain unchanged Operation: revision 1 dated 24/07/2006 of MAT Aviation's French flight ma (...) | Limited to aircraft EC120B S/N 1049, G-VIPR only, when equipped with EASA.R.S.01017 and EASA.IM.R.S.01056 | 04.08.2006 |
| 10017193 | AEROTEC CONCEPT | EASA.R.S.01330 | | | EUROCOPTER FRANCE SA 365N | AEROTEC avionics modification n° 6306: installation of two NAV/COM/GPS GARMIN GNS 430 for IFR B-RNAV "En route" operations, with their associated GARMIN GA 56 GPS antenna, two MID CONTINENT MD 41 Annunciation Control Units, an ALLIED SIGNAL KA 33 cool (...) | | Installation: AEROTEC STC Implementation File n°6306, revision 1 dated 26/10/06, or later revision. Operation: AEROTEC Flight Manual Supplement n°6306-AFMS, edition 1 revision 0, dated January 29th 2007 or later approved revision. Inspection and maintenance: Refer to individual equipment Manuals. | This STC is limited to SA 365 N s/n 6218 and can be installed only by AEROTEC technical staff in accordance with AEROTEC STC Implementation File n°6306, revision 1 dated 26/10/06, or later revision. AEROTEC minor modification n°6206, approved under (...) | 12.02.2007 |
| 10034668 | AEROTEC CONCEPT | | | | C-212-DD (C212-300) | Mod 0921: EGPWS Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-25 Amendment 1 The certificated noise and/ or emissions levels of the original produ (...) | Master Data List: AE0921MDL Rev. 7 dated 22. Mar.2011 Airplane Flight Manual supplement: AE0921-AFMS Rev. Draft 2 dated 19.Feb.2011 Applicable revisions or later revisions of the above listed documents approved by EASA | This STC is limited to the MSN 387 | 19.04.2011 |
| 10034675 | AEROTEC CONCEPT | | | | C90A | Installation d'une antenne SATCOM indium - dossier aerotec1082rev1 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 23 303, 305, 307, 365, 571, 573, 603, 605, 607, 609 The certificated noise and/ or em (...) | none | none | 19.04.2011 |
| 10034876 | AEROTEC CONCEPT | P-EASA.A.S.03826 | | | C-212-300 | INSTALLATION OF HONEYWELL CAS 67A TCAS II SYSTEM AND ATC TRANSPONDER MODE S HONEYWELL MST 67A IN THE AIRPLANE CASA C-212-300 S/N 387 | The Certification Basis for the original product remains applicable to this certificate/ approval | Technical Documents to be Approved by EASA: 1) Master Drawing List AE0906MDL Rev. 9, 19/04/11 2) Supplement to the AFM AE0906SMV Re.2, 17/03/11 3) Instructions for Continuous Airworthiness AE0906IPMN, Rev.3, 04/04/11 or later revisions of the above listed documents approved by EASA | The Airworthiness Approval is limited to the Airplane CASA C-212 S/N 387 | 09.05.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------|--------------|----------------|----------------------|---|--|--|---|---|---------------|
| 10034891 | AEROTEC CONCEPT | | | | A340-313 | Installation of EFB class II this STC installs the Rosemount Aerospace 8700C2-3 Laptop Docking Station (System provision) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: EASA TCDS EASA.A.015C.R1 F-2305-002- Installed Provisions for EFB Class 2 Provisions T (...) | Master Data List, ref: AE1098MDL, dated 09 May 2011 Aircraft Flight Manual Supplement, ref: AE1098SMV, dated 18 March 2011 Instruction for Continued Airworthiness, ref: AE1088IPMN, dated 17 March 2011 or later revisions of the above listed documents approved by EASA | None | 10.05.2011 |
| 10036025 | AEROTEC CONCEPT | | | | PA 30 | PFM ASPEN EFD 1000 PRO INSTALLATION on Piper PA 30. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List « INSTALLATION PFM ASPEN EFD 1000 PRO SUR PIPER PA 30 » Ref. AE 1010 MDL Rev. 5 dated 10/02/2011 Installation in accordance with SERVICE BULLETIN Ref. AE 1010 SB Rev. 0 dated 02/04/2011 Maintenance in accordance with the Instructions f (...) | In case of malfunction of the Aspen EFD 1000 Pro PFD or divergence of indications, use standard instruments for reference. AHSR reset prohibited in flight. LPV Approaches are not authorized. RSM GPS to be used only in case of emergency. | 05.08.2011 |
| 10036974 | AEROTEC CONCEPT | | | | C90A | Installation of Honeywell TCAS II ATC and GPWS MK VI according to Aerotec modification n° MOD 1085. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS_23 amdt 1: 301, 303, 305, 337, 561, 571, 603, 607, 867, 1351, 1353, 1357, 1365, 1367, (...) | Flight Manual Supplement n° AE1085SMV Revision 1 dated 21/03/2011. Instruction for continued airworthiness n° AE1085IPMN revision 1 dated 1/04/2011. Master Data List n° AE1085MDL revision 1 dated 22/03/2011 or later revisions of the above listed documents approved by EASA. | This change is applicable to aircraft s/n LJ1066. | 21.10.2011 |
| 10037027 | AEROTEC CONCEPT | | | | C90A | Installation of Garmin G600 EFIS and Garmin GNS 430 W COM/NAV/GPS system. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS23 amdt 1: 301, 303, 305, 337, 561, 571, 603, 607, 771, 773, 867, 1301, 1303, 1309, 1 (...) | Flight Manual supplement No. AE1084SMV revision 2 dated 21/06/2011. Instructions for Continued Airworthiness No. AE1084IPMN revision 1 dated 1/04/2011. Master Data List No. AE1084MDL revision 4 dated 21/10/2011 or later revisions of the above mentioned documents approved by EASA. | 1- This change is applicable to aircraft s/n LJ 1066 and its approved for BRNAV operations. 2- The airplane must be equipped with: - an altitude indicator, an airspeed indicator and an attitude independent from the G600 EFIS. - a third radio-communication or radio-navigation system, independent | 26.10.2011 |
| 10037808 | AEROTEC CONCEPT | | REV. 1 | | A319-115, -133 A320-214, -216 | Advanced Mission Display System - phase B (AMDS/B) Activation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: Electrical Wiring Interconnection System as per CS-25, Amdt 10, Subpart H Therequiremen (...) | - Certification Program, ref. doc: AE11043PCDP, Rev. 3, dated 16th November 2011 - Master Data List, ref. doc: AE11043MDL, Rev. 2, dated 1st December 2011 - Flight Manual Supplement, ref. doc: AE11043SMV, Rev. 0, dated 21st November 2011 or later revisions of the above listed documents approved by EASA | None | 22.12.2011 |
| 10038070 | AEROTEC CONCEPT | | REV. 1 | | A330-243/-301/-341/-343 | This modification intends to: - Install the Ball#Mount support on the cockpit sliding windows - Install the two Pilot Terminal Units (PTU/B) in the cockpit on their Ball#Mount support - Install OVCU thin mounting tray - Install AMPD 3MCU fixed mounting t (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: A330 EASA TCDS A.004 The requirements for environmental protection and the associated de (...) | Certification Programme (including Master Data List in chapter 9) AE11061PCD Instructions for Continued Airworthiness AE11061CA Flight Manual Supplement AE11061SMV or later revisions of the above listed documents approved by EASA | None | 05.06.2012 |
| 10038946 | AEROTEC CONCEPT | | | | PA-23-250 | Installation of Aspen EFD 1000 Pro electronic flight instrument system. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS23 amdt 1 The certificated noise and/or emissions level of the original product are unchanged (...) | Aerotec Flight Manual Supplement No. AE11143SMV revision 2 dated 07/03/2012. Aerotec Master Data List No. AE11143MDL revision 1 dated 22/03/2012. Aerotec ICA No. AE11143ICA revision 0 dated 28/10/2011 or later revisions of the above listed documents approved by EASA. | Initial installation of this design change must be performed by AEROTEC technical personnel. This installation is approved for BRNAV operations with coupling to S-TEC 55X autopilot. The modified airplane must be equipped with an attitude indicator, anal (...) | 30.03.2012 |
| 10039917 | AEROTEC CONCEPT | | | | 210N | Installation of a Garmin G500 EFIS, two Garmin GNS 430 W systems and associated GA 35 GPS antennas, a Garmin GTX 330 mode S transponder, a Garmin GMA 340 audiopanel, a Honeywell KN62A DME receiver, a Honeywell KR 87 ADF receiver and associated KA 44B antenna. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 23 The Certificated noise and/or emissions levels of the original product are unchanged (...) | Aerotec Flight Manual Supplement Nr. AE11094SMV revision 2 dated 21/05/2012. Aerotec Instructions for continued airworthiness Nr. AE11094IPMN revision 1 dated 10/02/2012. Aerotec Master Data List Nr. AE11094MDL revision 0 dated 22/05/2012 or later revis (...) | This STC is limited to aircraft s/n 62969. This avionics installation is approved for BRNAV operations with coupling to S-TEC 55X autopilot. The aircraft must be equipped with: back-up airspeed, altitude, attitude, magnetic heading and navigation indic (...) | 01.06.2012 |
| 10040456 | AEROTEC CONCEPT | | | | EC225LP | SYSTEME ANTI-GIRATOIRE/Resling load stabilization system | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 29 Issue 2 paragraphs: 301, 303, 305, 307, 337, 603, 610, 865, 1351, 1353, 1357, 1301 (...) | - Flight Manual Supplement EC 225 LP MOD AEROTEC CONCEPT No.1078 SYSTEME ANTIGIRATOIRE, document AE1078SMV rev 0 dated 02/05/2012 or later revisions of the above listed documents approved by EASA - INSTRUCTIONS FOR CONTINUED AIRWORTHINESS MOD 1078 EC22 (...) | S/N 2797 and 2824/VFR only Required pre-modification: Aerotec minor modification: AE11142MSSL (Logging EC225 LP) and use of the dedicated RFM "SELECTION DEUR DE MISSION: UPPER/LOWER (Transport de charge externe à #6 lingue)", EASA approval 10037964 | 04.07.2012 |
| 10043477 | AEROTEC CONCEPT | | | | A330-201/-202/-203/-223/-243 A330-301/-302/-303/-321/-322 A330-323/-341/-342/-343 | A330 Air China WiFi System The modification deals with the installation of a WiFi system on Air China A330 aircraft. The WiFi system is built with 2 LRJ installed in Electronic Bay (AMDP and PoE) and 7 WiFi Access Points dispatched in the passage cabin (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List AE12072MDL Rev.0, dated 23/01/2013. Flight manual supplement AE12072AFMS Rev. 0, dated 23/01/2013. Instructions for Continued Airworthiness: AE12072CA Rev. 0, dated 23/01/2013 or later revisions of the above listed documents approved by EASA. | None | 31.01.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------|-------------------------|----------------|----------------------|--|---|--|--|---|---------------|
| 10043783 | AEROTEC CONCEPT | | | | TB 20 | Installation of a Garmin GNS 530W system, GPS GA 35 antenna and a KindKN 62A DME | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amdt. 2 The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement No AE12087SMV/GNS 530W, revision 0 dated 02/08/2012 Installation Instructions No AE12087SB, revision 0 dated 02/08/2012 Instructions for Continued Airworthiness No AE12087PMN, revision 0 dated 02/08/2012 Master Data List No AE120 (...) | This installation is limited to airplane TB20 S/N 1088 and it is approved for IFR B-RNAV operations with coupling to a King KFC 150 autopilot and to an Aspen EFD 1000 Pro display. Software upgrades introducing new functions at equipment level must be the subject of (...) | 25.02.2013 |
| 10043786 | AEROTEC CONCEPT | | | | TB20 | Installation of an Aspen Evolution 2000 display system | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amdt. 2 The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement No AE12096SMV, revision 0 dated 17/09/2012 Installation Instructions No AE12096SB, revision 0 dated 17/09/2012 Instructions for Continued Airworthiness No AE12096CA, revision 0 dated 17/09/2012 Master Data List No AE12096MDL, revision 0 dated 17/09/2012 | The airplane must be equipped with standby altitude, airspeed, attitude, heading and navigation indicators independent from and not similar to the Aspen primary display. Software upgrades introducing new functions at equipment level must be the subject of (...) | 25.02.2013 |
| 10044029 | AEROTEC CONCEPT | | | | DR 400/500 | Installation of an EFIS Aspen EFD 1000 Pro and a Garmin GTN650 system with associated antennas, according to Aerotec modification n° AE12079. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement n° AE12079SMV EFD 1000, revision 0 dated 04/03/2013 Flight Manual Supplement n° AE12079SMV GTN 650, revision 0 dated 04/03/2013 Instructions for continued airworthiness n° AE12079ICA revision 1 dated 26/10/2012 Master Data List n° (...) | This STC is limited to airplane DR400/500 s/n 41. Initial installation on aircraft must be performed by authorised Aerotec technical personnel. The installation is approved for IFR BRNAV, PRNAV, RNP approach LNAV and LPV operations. LNAV/VNAV APV Baro sp (...) | 13.03.2013 |
| 10044859 | AEROTEC CONCEPT | | | | DC-9-83 (MD-83) | Cabin modification and refurbishing. Refurbishing of the cabin and addition of a VIP private salon; updating airshow system; replacement of the path-marking system by a fluorescent one. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special conditions: D-01 Glass screen of display-monitors; D-04 Isolated compartments; H-01 Enhanced airworthiness programme 1 (...) | Master Data List AE12056MDL Rev. 2 dated 08.05.2013 Master Drawing List AE12056MDWL Rev. 0 dated 06.05.2013 Airplane Flight Manual Supplement AE12056AFMS Rev. 1 dated 08.05.2013 Instruction for Continued Airworthiness AE12056ICA Rev. 2 dated 15.04.2013 (...) | DC-9-83 (MD-83) Serial Number 49809 | 10.05.2013 |
| 10046101 | AEROTEC CONCEPT | | | | MF50 | F50 220V/50Hz installation Description as per Associated Technical Documentation | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List AE13119MDL Rev. 1, dated 02/08/2013, Certification Program AE13119PCDP Rev. 2, dated 11/07/2013, Safety Analysis AE13119AS Rev. 0, dated 03/06/2013, Electrical Loads Analysis AE13119ELA Rev. 0, dated 03/06/2013, Structural Analysis AE131 (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 14.08.2013 |
| 10046517 | AEROTEC CONCEPT | | | | PA-30 | STC 13099 - installation GTN650 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: Refer to AE13099PCDP Prev. 1 for paragraphs at CS23 amdt3. The requirements (...) | AE130999ICA Rev 0 dated 07/05/113 - Instructions for Continued Airworthiness AE13099MDL Rev 2 dated 24/09/13 - Master Drawing List AE13099SMV Rev 0 dated 29/08/13 # Airplane Flight Manual Supplement or later revisions of the above listed documents approved by EASA. | Limitation to PA-30 serial number 301251 and 301582. | 27.09.2013 |
| 10017251 | AEROTEC SARL | EASA.R.S.01452 | REV. 1 | | SA365N | AEROTEC modification IRB (installation radio de bord) n° 0928. (Note for internal record: This application is to amend an existing EASASTC (Ref P-EASA.R.S.01452) to add the S/N 6147) | FAR 29 amendment 16 The Certification Basis for the original product stated in TCDS N.159 remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation - Aerotec Master Data List AE 0809 MDL Rev. 4 dated 23/03/09 - Aerotec Master Data List AE 0928 MDL Rev. 0 dated 23/02/10 - Operation - Aerotec Flight Manual Supplement AE 0809 AFMS Rev. 1, dated 23/03/09 or later approved revision - Aerotec Fligh (...) | This STC is limited to SA 365N S/N 6080 <(-> S/N 6147 and can be installed only by AEROTEC technical staff in accordance with -AEROTEC STC file Mod n° 0809, for S/N 6080-AEROTEC STC file Mod n° 0928, for S/N 6147 | 01.03.2010 |
| 10043086 | AEROTEC SARL | EASA.A.S.03701 | | | TB 20 | Installation of L3 Communication Stormscope WX500, Garmin audio system GMA 340 and Garmin GNS 530 W for IFR B-RNAV enroute navigation. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amdt. 2 The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - , Aerotec Flight Manual supplement no. AE0841SMV, revision 0, dated 04/02/09 - , Aerotec Instructions for Continued Airworthiness no. AE0841IPMN, revision 0, dated 04/02/09 - , Aerotec Master Data List no. AE 0841 MDL, revision 1, dated 02/03/09 | This installation is limited to aircraft TB 20 s/n 1801 which must be equipped with a third radio-communication or radio-navigation means independent from GNS 430 and GNS 530 equipment. | 07.05.2009 |
| 10014279 | AEROTECHNIC GMBH | EASA.A.S.01516 | | | BEECH 200 BEECH 300 BEECH 300LW BEECH B200 BEECH B300 | Installation of Honeywell CAS-67A ACAS-System and Dual MST 67A Dual Mode S Transponder System Enhanced Surveillance (Exchange of existing transponder) | | EASA Approved Airplane Flight Manual Supplement B300-123667-0601, Revision 1, dated 20.05.2006 or later EASA approved revision. | | 13.06.2006 |
| 10014280 | AEROTECHNIC GMBH | EASA.A.S.01517 | | | BEECH 200 BEECH 200C BEECH 300 BEECH 300LW BEECH B200 BEECH B200C BEECH B300 | Installation of Dual Mode-S Transponder System Rockwell-Collins TDR-94D and Collins CTL-92E | | EASA Approved Airplane Flight Manual Supplement 123567-0601, Revision 1, dated 06.09.2006 or later approved revision | | 16.11.2006 |
| 10014474 | AEROTECHNIC GMBH | EASA.A.S.02536 REV 1 | REV 1 | | F27 MK 500 | Change of WXR to RDR 2000 Radar | | Data package containing the following documents, and additional supporting data: 1. F27-12412302-0103 Rev. 0 Master Document list 2. F27-12412302-0101 Rev. 0 Compliance Program 3. F27-12412302-0102 Compliance Report/Checklist 4. F27-124123 (...) | | 04.08.2006 |
| 10014631 | AEROTECHNIC GMBH | EASA.A.S.02773 | | | DORNIER DO 228-212 | Installation of mission equipment for maritime surveillance. | | Compliance Programme DO228-12256201-0101 Rev. 2 Master Document List DO228-12256201-0109 Rev. 0 AFM-Supplement DO228-12256201-0601 Rev.1 | Restricted to S/N 8181, 8183. RUAG changes EASA.A.C.04131 and EASA.A.C.04132 must be installed concurrently. | 26.07.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|-------------------|----------------|----------------------|--|--|---|---|---|---------------|
| 10014693 | AEROTECHNIC GMBH | EASA.A.S.02858 | | | BEECH 200 BEECH 200C BEECH 300 BEECH 300LW BEECH B200 BEECH B200C BEECH B300 | Installation of Honeywell MK VI EGPWS and Avidyne TAS-620 TCAD. | | To be installed in accordance with Aerotechnic modification document ref 125204-01-1001 Rev 0 dated 15.11.2006 and 125204-01-1002 Rev 0 dated 10.10.2006, or later approved revision. To be operated in accordance with Aerotechnic Flight Manual Supplement (...) | | 05.12.2006 |
| 10015017 | AEROTECHNIC GMBH | EASA.A.S.03323 | | | B90, C90, C90A, C9GT | Installation of AVIDYNE TAS-620 TCAD and EX500 Display, together with necessary antennas | | EASA approved Compliance Programme, Doc. No. 126635-0101, Revision 1, dated 13-Dec-2007, or later approved revision EASA Approved Airplane Flight Manual Supplement, Doc. No. 126635-01-0601, Revision 1, dated 14-Dec-2007, or later approved revision Inst (...) | Only airplanes complying with the limitations and conditions listed in Chapter 2.1. #Limitations and Conditions for Installation#, of the Installation Manual, Doc No. 126635-01-1001, Revision 2, dated 12 Dec-2007, or later issued revision (...) | 19.12.2007 |
| 10043016 | AEROTECHNIC GMBH | EASA.R.S.00492 | | | BELL 430 | Installation of Cockpit Voice Recorder Universal CVR-120 | | - Bell 430-03-0101 or later EASA approved Revision - Flight Manual Supplement Doc. No. Bell 430-03-0601 Revision 0 or later EASA approved Revision. | 1., This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 10.01.2004 |
| 10043107 | AEROTECHNIC GMBH | EASA.A.S.02536 | | | F27 MK500 FRIENDSHIP | Change of WXR to RDR 2000 Radar | | Data package containing the following documents: 1. Fokker Services SB F27/28-70; Fuel - Introduction of Fuel ALis and CDCCLs for the F27/Mk200/300/400/500/600/700.2. Fokker Services Report SE-671: Fokker 50/60 Fuel ALis and CDCCLs. These reports are base (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall (...) | 04.08.2006 |
| 10014062 | AEROTECHNIC VERTRIEB- UND SERVICE | EASA.A.S.01148 | | | CESSNA 650 CITATION III | Installation of Honeywell CAS-67 ACAS II and Dual MST 67A Mode-S Transponder. | | - Airplane Flight Manual Supplement for Cessna 650 Citation III, Doc. No C650-01-0601 Revision 0, dated 17.03.2005 or later approved revision. - Instructions for Continuous Airworthiness, Doc. No C650-01-0701, Revision 0, dated 17.03.2005 or later approved revision. | | 11.04.2005 |
| 10014063 | AEROTECHNIC VERTRIEB- UND SERVICE | EASA.A.S.01149 | | | CESSNA 650 CITATION III | Installation of Allied Signal Solid State Flight Data Recorder. | | - Airplane Flight Manual Supplement for Cessna 650 Citation III, Doc. No C650-02-0601 Revision 0, dated 02.03.2005 or later approved revision. - Instructions for Continuous Airworthiness, Doc. No C650-02-0701, Revision 0, dated 02.03.2005 or later approved revision. | | 11.04.2005 |
| 10014064 | AEROTECHNIC VERTRIEB- UND SERVICE | EASA.A.S.01150 | | | CESSNA 650 CITATION III | Installation of Allied Signal AR120 Solid State Cockpit Voice Recorder. | | - Airplane Flight Manual Supplement for Cessna 650 Citation III, Doc. No C650-03-0601 Revision 0, dated 12.03.2005 or later approved revision. - Instructions for Continuous Airworthiness, Doc. No C650-03-0701, Revision 0, dated 12.03.2005 or later approved revision. | | 11.04.2005 |
| 10014098 | AEROTECHNIC VERTRIEB- UND SERVICE | EASA.A.S.01199 | | | CESSNA 550, CESSNA 560 | Installation of Allied Signal SSUFDR | | Master Document List C550-122401-0107 Rev. 0 or later approved Revision Flight Manual Supplement C550-122401-0601 Rev. 0 or later approved Revision | Certification Basis of the Change: FAR 25 as defined in FAATCDS A22CE Rev. 44 | 19.04.2006 |
| 10014549 | AEROTECHNIC VERTRIEB- UND SERVICE | EASA.A.S.02639 | | | F27 MARK 500 | Installation of Smoke Detection System | | Master Document List MDL F27-12412308-0109 Revision 1, dated 23.06.2006 Airplane Flight Manual Supplement F27-12412308-0601 Declaration of Compliance of Aerotechnic GmbH EASA.21J.099, dated 11.08.2006 | FAR-25 Amdt. 119 | 14.08.2006 |
| 10014829 | AEROTECHNIC VERTRIEB- UND SERVICE | EASA.A.S.03042 | | | 550 BRAVO, 560, 560 XL CESSNA 500, 550, S550, | Installation of Honeywell MK VIII EGPWS. | | Master Document List 125706-0109, Issue 1, or later approved revision. Instructions for Continued Airworthiness 125706-0701, Rev 0, dated 02.08.2007, or later approved revision. AFM Supplement 125706-0601, Rev 1, dated 17.08.2007, or later approved revision. | | 07.09.2007 |
| 10017106 | AEROTECHNIC VERTRIEB- UND SERVICE | EASA.R.S.01081 | | | MD900 | Installation of a Standby Altitude Indicator and a KANNAD 406 ELT in a MDH1 MD900 | | Flight Manual Supplement S1by Altitude Gyro, MD900-01-0601 Rev.1, Flight Manual Supplement MD900 KANNAD 406 AF-H ELT MD900-01-0602 Rev:1 | Installation is restricted to "VFR only" | 22.06.2005 |
| 10044596 | AERSOLUTION GmbH | | | | AW109SP | Quick change patient evacuation system - The installation is identified by the p/n 109INS-2520001. It consists in an interior modification with EMS. | The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | 109AST-FMS-001-A RFM Supplement or later revisions of the above listed documents approved by EASA. 109AST-CPL-001-E Certification Plan 109AST-DOC-001 Declaration of Compliance (dated 18 April 2013) 109AST-MDL-001-D Master Document List 109AST-ICA-001-A Operation & Maintenance Manual | The compatibility of the "Quick Change Patient Evacuation System" with other previously approved kit must be determined by the installer. Prior to installation of this design change it must be determined that the interrelationship between this design c (...) | 22.04.2013 |
| 10016766 | AGUSTAWESTLAND PHILADELPHIA | EASA.IM.R.S.01097 | | TCCA STC SH24NE | AGUSTA A-109 AGUSTA A-109 All | Installation of Alternate Static Pressure Source | | Configuration A: in accordance with Agusta drawings 109-A720-14 and 109-A310-55-77 all dated 27-7-84. Configuration B: Agusta kit No. 109-A720-14, in accordance with Agusta Service Instruction No. 109-A-005 dated 11 February 1985. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. Configuration (...) | 04.07.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|----------------|----------------|----------------------|------------------|--|---|---|---|---------------|
| 10031050 | AGUSTAWESTLAND PHILADELPHIA | | | FAA STC SR02767NY | AW119MKII | Installation of an Artex ME406P ELT | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Agusta Aerospace Corporation installation master drawing list, report no. AAC-ENG-09-031, rev - , dated 08 October 2009- Agusta Aerospace Corporation Instructions for Continued Airworthiness report no. AAC-ENG-09-038, rev - , dated 7 October 2009 (...) | none | 23.07.2010 |
| 10031114 | AGUSTAWESTLAND PHILADELPHIA | | | FAA STC SR02777NY | AW119MKII | Installation of a SX-5 Searchlight and cargo Hook Heated rearview mirror | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Agusta Aerospace Corporation installation master drawing list, report no. AAC-ENG-09-041, rev A, , dated 20 December 2009- Agusta Aerospace Corporation Rotorcraft Flight Manual, report no. AAC-ENG-09-048, rev - , dated FAA approved March 15, 20 (...) | none | 28.07.2010 |
| 10031274 | AGUSTAWESTLAND PHILADELPHIA | | | FAA STC SR02265NY | A119/AW119MKII | Installation of pedal extensions on A119, AW119MKII helicopters | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | For A119 helicopter:- Agusta Aerospace Corporation installation master data list, report no. AAC-ENG-05-091, rev - , dated 24 April 2006. For AW119MKII helicopter:- Agusta Aerospace Corporation installation master data list, report no. AAC-ENG-08-086, r (...) | none | 06.08.2010 |
| 10031277 | AGUSTAWESTLAND PHILADELPHIA | | | FAA STC SR02789NY | AW119MKII | Installation of a Border Guard Configuration | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Agusta Aerospace Corporation installation master data list, report no. AAC-ENG-09-011, rev A, , dated 2 February 2010- Agusta Aerospace Corporation Rotorcraft Flight Manual, report no. AAC-ENG-09-018, rev - , dated 02 April 2010 or later revisions of (...) | none | 06.08.2010 |
| 10031316 | AGUSTAWESTLAND PHILADELPHIA | | | FAA STC SR02428NY | A119 / AW119MKII | EASA Validation FAA STC SR02428NY for A119 / AW119MKII - High Visibility Crew Doors | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirement for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Instructions for the High Visibility Crew Doors, Report n. AA-04051, Revision F, dated March 27, 2007 for the A119. Installation Instructions for the High Visibility Crew Doors, Report n. AAC-ENG-08-075, Revision - , dated November 12, 2008 (...) | None | 09.08.2010 |
| 10039055 | AGUSTAWESTLAND PHILADELPHIA | | | FAA STC SR02444NY | A119 AW119 MKII | EASA Validation of FAA STC SR02444NY "Installation of the Offshore Configuration in accordance with Agusta Aerospace Master Data List AAC-ENG-06-051 Rev. C" | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Agusta Aerospace Master Data List AAC-ENG-06-051 Rev. C (-,-) dated March 27, 2008. Agusta Aerospace Instructions for Continued Airworthiness for the Offshore Configuration, Report No. AAC-ENG-06-057 Rev. A dated November 14, 2007. Agusta Aerospace Roto (...) | None | 03.04.2012 |
| 10044017 | AGUSTAWESTLAND PHILADELPHIA | | | FAA STC SR02135NY | A119/AW119MKII | Pilot and Crew Doors removed | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Rotorcraft Flight Manual Supplement Reports No. AAC-ENG-04-046 Rev. B (FAA approved on June 30, 2005) for the A119 model AAC-ENG-08-084 Rev. - (FAA approved on March 4, 2009) for the AW119MKII model or later revisions of the above listed documents app (...) | None | 08.04.2013 |
| 10014165 | AIM AVIATION (HENSHALLS) LTD | EASA.A.S.01297 | | | AIRBUS A300-600 | Installation of a Galley bulkhead. | | AIM Aviation (Henshalls) Definition Document DD 1011 Issue New, dated 12, April 2005 or later EASA approved revisions. | 1. Viewing of all Placards and signage to be in compliance with CS/FAR 25.791 Weight, Centre Gravity and Weight Distribution to be in Compliance with CS/FAR 25.1519.2. Compliance with CS/FAR 25.785 (d) (3), the elimination of any injurious subject wit (...) | 08.06.2005 |
| 10014166 | AIM AVIATION (HENSHALLS) LTD | EASA.A.S.01299 | | | BOEING 767-300 | Installation of Stowage C2. | | AIM Aviation (Henshalls) Definition Document DD 1009 Issue New, dated April 5, 2005 or later EASA approved revisions. | 1. Viewing of all Placards and signage to be in compliance with CS/FAR 25.791 Weight, Centre Gravity and Weight Distribution to be in Compliance with CS/FAR 25.1519.2. Compliance with CS/FAR 25.785 (d) (3), the elimination of any injurious subject wit (...) | 08.06.2005 |
| 10014167 | AIM AVIATION (HENSHALLS) LTD | EASA.A.S.01300 | | | BOEING 737-500 | Installation of Galley 3. | | AIM Aviation (Henshalls) Definition Document DD 1010 Issue New dated April 11, 2005 or later EASA approved revisions. | 1. Viewing of all Placards and signage to be in compliance with CS/FAR 25.791 Weight, Centre Gravity and Weight Distribution to be in Compliance with CS/FAR 25.1519.2. Compliance with CS/FAR 25.785 (d) (3), the elimination of any injurious subject wit (...) | 14.06.2005 |
| 10014168 | AIM AVIATION (HENSHALLS) LTD | EASA.A.S.01301 | | | BOEING 737-500 | Installation of Wardrobe W1. | | AIM Aviation (Henshalls) Definition Document DD 1015 Issue New, dated April 25, 2005 or later EASA approved revisions. | 1. Viewing of all Placards and signage to be in compliance with CS/FAR 25.791 Weight, Centre Gravity and Weight Distribution to be in Compliance with CS/FAR 25.1519.2. Compliance with CS/FAR 25.785 (d) (3), the elimination of any injurious subject wit (...) | 08.06.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|----------------|----------------|-----------------------|------------------------------------|--|---|--|--|---------------|
| 10014356 | AIM AVIATION (HENSHALLS) LTD | EASA.A.S.01621 | | | BOEING 737-400 | Installation of Galley 1 & Galley 2 for Futura International Airways | | DD 1027: issue new or later approved issue | | 06.01.2006 |
| 10014453 | AIM AVIATION (HENSHALLS) LTD | EASA.A.S.02507 | | | BOEING 747-428 | Installation of Upper Deck Galley partnr 747FA01GA00 into a B747-428 passenger to cargo conversion (747-400BCF) as defined in STC Definition Document DD 1022 d.d. 01-06-2006 or later EASA approved revisions. | | - AIM Henshalls Component Maintenance Manual 25-36-76- AIM Henshalls Illustrated Parts List 25-36-76 | | 03.05.2007 |
| 10014479 | AIM AVIATION (HENSHALLS) LTD | EASA.A.S.02542 | | | BOEING 767-300 | Installation of Galley F4 Part Number 7673FCF4GA00 | | AIM Aviation (Henshalls) Change Number HEN 1035 | Restricted to Aircraft Serial Number 26063 & 26064 | 29.09.2006 |
| 10015007 | AIM AVIATION (HENSHALLS) LTD | EASA.A.S.03307 | | | BOEING 757-200 | Installation of Crew Galley in B757-200 Freighter Conversion. | | AIM Aviation (Henshalls) STC Definition Document, DD 1090, Issue New, 24/10/2007 Instruction for continued Airworthiness, ICA/PR/7001, Rev. New, Nov 2007. | | 05.02.2008 |
| 10034166 | AIM AVIATION (HENSHALLS) LTD | | | | B757-200 | Installation of a Crew Galley in the G1A Position on a Freighter Conversion a/c | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this approval. | AIM Aviation (Henshalls) STC Definition Document, DD 1189, Issue New, 04/02/2011. Instructions for Continued Airworthiness, ICA/PR/7002, Rev New, Feb 2011. or later revisions of the above listed documents approved by EASA | None | 14.03.2011 |
| 10036493 | AIM AVIATION (HENSHALLS) LTD | | | | A319 | Installation of a Galley 13 and Closet C1 into an Airbus A319-100 to be operated by BMI | CS 25 Amendment 11 | Change Record HEN 1208 Issue 1 for later revisions of the above listed documents approved by EASA | None | 13.09.2011 |
| 10038683 | AIM AVIATION (HENSHALLS) LTD | | | | A321 | Installation of Galley 2A Part No. 321BM2AGA00 and Galley 3A Part No 321BMT3AGA00 into an A321-200 to be operated by BMI. | CS 25 Amendment 11 The Certification Basis for the original product remains applicable to this certificate/ approval. | Change Record HEN 1207, Issue 1 | None | 07.03.2012 |
| 10040427 | AIM AVIATION (HENSHALLS) LTD | | | | B747-400 | Installation of Galley 2A, 2B, 4B, 4C and 4D and ceilings in the G4 complex in a Boeing 747-400 to be operated by Virgin Atlantic Airways in accordance with AIM Henshalls STC Definition Document DD 1198<-,->dated 3 January 2012 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | EASA Supplemental Type Certificate Definition Document DD 1198<-,->dated 3 January 2012 or later revisions of the above listed documents approved by EASA | None | 02.07.2012 |
| 10044656 | AIM AVIATION (HENSHALLS) LTD | | | | A330-300 | Installation of a Galley and associated ceiling panels into an Airbus A330-300 to be operated by Delta. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AIM Aviation Ltd. Definition Document No. DD 1209 Issue New Date 28/03/13 or later revisions of the above listed documents approved by EASA | None | 26.04.2013 |
| 10045392 | AIM AVIATION (HENSHALLS) LTD | | | | A330-200 | Installation of Galleys and associated ceiling panels into an Airbus A330-200 to be operated by Delta. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AIM Aviation Ltd Definition Document No. DD1210 Issue New Dated 14 May 2013 or later revisions of the above listed documents approved by EASA, | None | 20.06.2013 |
| 10014629 | AIM AVIATION (JECCO) LIMITED | EASA.A.S.02768 | | | AIRBUS A330-200 AIRBUS A330-300 | Introduction of flight crew rest facility into A330-200 and -300 aircraft. | | Major change MM/334/06.SB AIMA330-25-004 | MM-358-07 must be installed. | 31.05.2007 |
| 10014649 | AIM AVIATION (JECCO) LIMITED | EASA.A.S.02804 | | | AIRBUS A330-200 AIRBUS A330-300 | Aer Lingus Flight Crew Rest Facility | | Change No. MM/331/06 Operation & Maintenance Manual: Publication 1/07CMM/ AIM Publication 25-70-03 | | 05.02.2007 |
| 10014919 | AIM AVIATION (JECCO) LIMITED | EASA.A.S.03165 | | JAA/25/03/028 ISSUE 1 | BOEING 747-400 | Virgin Atlantic Airways B747-400 CoCo Reconfiguration. | | Major change MM/373-07 Issue A and Appendix 1 Issue C, and, Master Drawing List DR1281-07, Issue 2, or later approved revisions. | The following EASA approved STC's must be installed concurrently with this design change: EASA.A.S.02860 EASA.I.M.A.S.02572 | 09.10.2007 |
| 10015102 | AIM AVIATION (JECCO) LIMITED | EASA.A.S.03445 | | | | Premium Cabins Upgrade Appearance, comfort and maintenance improvement changes to the seats and the window reveals and shades in the First and Business Class cabins. | | AIM Major Change, ref. Doc. No. MM385/07, Issue C, dated 1st July 2008, AIM Certification Plan and Compliance Check List, ref. Doc. Appendix to Major Change MM385/07, Issue C, dated 1st July 2008, Declaration of Compliance - page 55 of th (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 05.08.2008 |
| 10015103 | AIM AVIATION (JECCO) LIMITED | EASA.A.S.03446 | REV. 1 | | AIRBUS A330-243 | Major Change to STC EASA.A.S.03446 (AIM Major Change MM384-07) date of issue 14 November 2008. Introduce decorative finish change to Contour 0995 series Business Class Seats. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. EASA Special condition for seat heat release requirements (CRID-4322-001) for areas (...) | AIM MAJOR Change ref. Doc. MM481-10 Issue C dated 21 December 2010. - AIM Certification and Compliance Check List, ref. Doc. Appendix to MM481-10 Issue C dated 21 December 2010. - Declaration of compliance, page 24 of AIM Certification and Compliance C (...) | EASA Supplemental Type Certificate EASA.A.S.03446 (AIM Major Change MM384-07) must be embodied prior to the installation of this Major Change. As per indications stated in AIM MAJOR Change ref. Doc. MM481-10 Issue C. | 22.12.2010 |
| 10040431 | AIM AVIATION (JECCO) LIMITED | | | | 747-400 | Installation of a single isolation switch in accordance with AIM Aviation Service Bulletin B747-24-011 dated April 23, 2012 as replacement for AIM Aviation Service Bulletin B747-24-04 or B747-24-05 in order to comply to FAA AD 2009-15-12. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Service Bulletin B747-24-011 dated April 23, 2012 Flight Manual Supplement Document Nr 11184 Issue A, dated 28 May<-,->2012 | none | 03.07.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------|-------------------|----------------|----------------------|--|--|---------------------|---|--|---------------|
| 10015599 | AIM AVIATION INC. | EASA.IM.A.S.01535 | | ST01504SE | BOEING 757-200 | Installation of a Flight Deck Galley | | AIM Aviation Inc. Master Drawing List AIM-MDL-14492 Revision New dated January 25 2005 or later approved revision | | 03.04.2006 |
| 10015600 | AIM AVIATION INC. | EASA.IM.A.S.01536 | | FAA STC 101490SE | BOEING 757-200 SERIES | Installation of a Flight Deck Lavatory | | AIM Aviation Inc. Master Data List AIM-DL-14486 Revision A dated December 17 2004 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 31.03.2006 |
| 10014775 | AIR ALLIANCE GmbH | EASA.A.S.02973 | | | SOCATA TBM 700 | Installation of Garmin GTX330D Transponder with Top Antenna. | | # Installation and Maintenance Manual: 2006_002_2070_500Rev. 0# Flight Manual Supplement: 2006_002_2070_510 Rev.2 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 15.08.2007 |
| 10014857 | AIR ALLIANCE GmbH | EASA.A.S.03081 | REV. 3 | | CESSNA 208 CESSNA 208A CESSNA 208B | Conversion of airplane for parachute operations.Installation of PARAKIT NAY001 or ParaKit C208-83 00 00 "System Gomolzig". | | For PARAKIT NAY001# Master Data List. PARAKIT NAY001 Model 208B, Revision 0, dated 20-Mar-1997 or later approved issue.# LBA approved Pilots Operating Handbook Supplement, Doc. No. PARAKIT NAY001, Revision 0, dated 20-Mar-1997, or later approv (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 29.01.2009 |
| 10014947 | AIR ALLIANCE GmbH | EASA.A.S.03207 | | | PIPER PA-42 | Installation of TCAS 9900 BX. | | Installation Instruction 2007_009_2063_500 Rev. 0Air Alliance - Einbau Avidyne TAS 620 (9900BX) TCAD System AFM supplement. | | 06.12.2007 |
| 10014988 | AIR ALLIANCE GmbH | EASA.A.S.03281 | | | PIPER PA-31T1 | Installation of Avionic Upgrade according MDL 2007_019_2029AA Form 1 Rev. 0. | | Installation Instruction 2007_019_2039_500, Rev. 2Air Alliance AFM-S Doc. No 2007_019_2039_510, Original issue from 6. November 2007, or later EASA approved RevisionAir Alliance AFM-S Doc. No 2007_019_2039_511, Original issue from 6. November 2007, (...) | | 10.01.2008 |
| 10015178 | AIR ALLIANCE GmbH | EASA.A.S.03556 | REV. 1 | | BEECH 300 | Installation of L3 FDR1000; Collins EGPWS Mark VI; Honeywell CAS 67 & Dual MST 67A Mode S Diversity Transponder; Artex 406-2 ELT | | The installation of the equipment shall be done in accordance with Air Alliance Engineering Order 2008_014_A24CE_0500, Revision 00, Issued 20.06.2008, or later EASA approved revision. Operation in accordance with the applicable AFMS as per Air Allianc (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 20.06.2008 |
| 10015205 | AIR ALLIANCE GmbH | EASA.A.S.03606 | | | CESSNA 525 CESSNA 525A | Exchange of GPS Navigation System KING KLN 900 against Flight Management System Universal UNS 1Lw | | EASA approved Master Document List, Doc. No. 2008_034_A1W1 AA Form 2, Revision 1, dated 11-Aug-2008, or later approved RevisionEASA approved Airplane Flight Manual Supplement, Doc. No. 2008_034_A1W1_0506, Revision 1, dated 07-Aug-2008 or later appro (...) | 1. This STC is only eligible for aeroplanes equipped with GPS Receiver System KLN 900 and Collins Pro Line 21 Avionic System2.Prior to installation of this modification the installer must determine that the interrelationship between this m (...) | 12.08.2008 |
| 10015226 | AIR ALLIANCE GmbH | EASA.A.S.03643 | REV. 1 | | CESSNA 525 | Installation of Dual GTX 330D as Replacement of Dual KT 70 or KT 71 | | AirAlliance Master Document List 2008_036_A1W1AA Form 2, in latest approved RevisionTo be operated in accordance with AFM Supplement Document No 2008_036_A1W1_506, Rev 0, or later approved RevisionsTo be maintained i.a.w. ICA Document No 2008_036_A1W1_507, Rev 1, or later approved Revisions | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 05.09.2008 |
| 10015263 | AIR ALLIANCE GmbH | EASA.A.S.03721 | | | DORNIER DO 228-101 DORNIER DO 228-201 DORNIER DO 228-202 | Installation of FMS Universal UNS-1Lw. | | - Engineering order: 2008_055_2031B_0500, Rev. 00- Instructions for Continued Airworthiness: 2008_055_2031B_0507, Rev. 01- AFM Supplements: 2008_055_2031B_0506, Rev. 03 | 1. Restricted to aircrafts which have already installed a LH HSI Honeywell KPI552B and two Shadin ADC#s ADC2000.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will i | 23.01.2009 |
| 10015271 | AIR ALLIANCE GmbH | EASA.A.S.03753 | | | PIPER PA-31T PIPER PA-31T1 PIPER PA-31T2 | Camera Installation Floor Cut Out | | According to AirAlliance Master Document List Doc. No. 2008_058_A8EA AA Form 2, Rev. 0, or later approved Revisions andAirAllianceInstallation Instructions 2008_058_A8EA_500, Rev. 0, or later RevisionsTo be operated in accordance with AirAlliance AF (...) | 1 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 21.01.2009 |
| 10015298 | AIR ALLIANCE GmbH | EASA.A.S.03851 | | | BEECH B300 BEECH B300C | Installation of Honeywell Mark VI EGPWS. | | - According to AirAlliance Master Document List Doc. No.2009_006_A24CE AA Form 2, Rev. 0, or later approved Revisions and AirAlliance Installation Instructions 2009_006_A24CE_500, Rev. 0, or later Revisions.- To be operated in accordance with (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 01.04.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------|-------------------|----------------|----------------------|--|--|---|--|--|---------------|
| 10015299 | AIR ALLIANCE GmbH | EASA.A.S.03852 | | | PIPER PA-31T PIPER PA-31T1 PIPER PA-31T2 | Installation/ Exchange of Avionics acc. AirAlliance MDL 2008_059_A8EA in latest approved Revision | | According to AirAlliance Master Document List Doc. No. 2008_059_A8EA Rev. 0, or later approved Revisions and AirAlliance Engineering Order 2008_059_A8EA-0500, Rev. 0, or later Revisions To be operated in accordance with AirAlliance AFM-S Document (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 01.04.2009 |
| 10016091 | AIR ALLIANCE GmbH | EASA.IM.A.S.02278 | REV. 3 | | PA-42-720 | Extension of pressurized fuselage life limit to account for the change in operational usage from Alitalia Airlines Training to Normal Usage by any operator. Note: This STC supersedes STC EASA.IM.A.S.02278 Rev 2 and indicates the transfer of an applicable S/N to a new STC holdership by AirAlliance. | The Certification Basis for the original product remains applicable to this certificate/ approval. FAR 23 Amtdt 1 to 16 The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are (...) | Air Alliance GmbH Report No AARD-1, Rev A dated 28 Dec 2006 AirAlliance PA-42-720 Aircraft - Fuselage Life Substantiation Air Alliance GmbH Report no AARD-2, Rev B dated 28 Dec 2006 AirAlliance PA-42-720 Aircraft - Pressurized Fuselage Cabin Instru (...) | This STC is limited only to airplane S/N 42-5501046 and 42-5501047 operated by any operator. Operation in normal usage up to a life-limit of 21,452 hours for S/N 42-5501046 and 21,561 hours for S/N 42-5501047. Air Alliance GmbH Report No AARD-2 Rev B dat (...) | 19.07.2011 |
| 10027217 | AIR ALLIANCE GmbH | P-EASA.A.S.03905 | REV. 1 | | TB9, TB10, TB 20, TB21, TB200 | Installation of Aspen EFD 1000 PRO | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CS-23 The certified noise and/ or emissions levels of the original product are unchanged (...) | Engineering Order: 2009_039_A51EU_0500, Rev. 0-Aircraft Flight Manual Supplement: 2009_039_A51EU_0506, Rev. 0-Instructions for Continued Airworthiness: 2009_039_A51EU_0507, Rev. 0 or later revisions of the above listed documents approved by EASA | None | 12.04.2010 |
| 10027714 | AIR ALLIANCE GmbH | P-EASA.A.S.03957 | | | TBM 700 | Installation of GPS Antenna for Situational Awareness Tool. | The Certification Basis for the original product remains applicable to this certificate/ approval. | Engineering Work Order 2009_025_A.010_500 Rev. 0; Instructions for Continued Airworthiness 2009_025_A.101.507 Rev. 0 or later revisions of the above listed documents approved by EASA | None. | 26.10.2009 |
| 10029583 | AIR ALLIANCE GmbH | P-EASA.A.S.03461 | | | LEARJET 35A | Installation of new avionics (FDR, TAWS, FMS, RDR/KMD) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval TGL No. 10 (P-RNAV) TGL No. 12 (TAWS) AMC 20-4 (B-RNAV) TGM 21/07 (Electrical Wiring) The c (...) | Refer to Master Document List 2008_006_A10CE AA Form 2 (Rev. 7) or later revisions of the above listed documents approved by EASA | Approval valid for s/n 35A-616 only | 02.04.2010 |
| 10033818 | AIR ALLIANCE GmbH | | | | PA-42-1000 | Installation of SAT antenna (Data-Link) into pressurized fuselage. | The Certification Basis for the original product and the following additional/ alternative airworthiness requirements are applicable to this change. For the areas affected by the change CS-23 has been elected to comply with where possible. The certified (...) | EASA approved Master Document List, Doc. No. 2010_040_PA42_AA Form 2 Master Doc List, Revision 1, approved 10-Jan-2011 Engineering Order, Doc. No. 2010_040_PA42_500, Rev. 1, released 24-Nov-2010 Instructions for continued Airworthiness, Doc. No. 2010_040 (...) | None. | 10.02.2011 |
| 10035779 | AIR ALLIANCE GmbH | | | | PA-42-720 | Extension of pressurized fuselage life limit to account for the change in operational usage from Alitalia Airlines Training to Normal Usage by any operator. Note: This STC indicates the transfer of an applicable S/N from STC EASA.IM.A.S.02278 Rev 2 to this STC holdership by AirAlliance | The Certification Basis for the original product remains applicable to this certificate/ approval. FAR 23 Amtdt 1 to 16 The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are (...) | Air Alliance GmbH Report No AARD-1, Rev A dated 28 Dec 2006 AirAlliance PA-42-720 Aircraft - Fuselage Life Substantiation Air Alliance GmbH Report no AARD-2, Rev B dated 28 Dec 2006 AirAlliance PA-42-720 Aircraft - Pressurized Fuselage Cabin Instru (...) | This STC is limited only to airplane S/N 42-5501045 operated by any operator. Operation in normal usage up to a life-limit of 21,563 hours for S/N 42-5501045. Air Alliance GmbH Report No AARD-2 Rev B dated 28 Dec 2006 Air Alliance PA-42-720 Aircraft - Pr (...) | 19.07.2011 |
| 10035987 | AIR ALLIANCE GmbH | | | | LEARJET 35A | Installation of Honeywell GNS-XLS second FMS system. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Air Alliance Master Document List 2011_006_LJ35 AA Form 2 Rev. 0; Aeroplane Flight Manual Supplement No. AA 2011_006_LJ35_0506 dated May 13, 2011 or later revisions of the above listed documents approved by EASA. | Limited to 35A-421 and 35A-616. | 03.08.2011 |
| 10037809 | AIR ALLIANCE GmbH | | | | 337, A, B, C, D, E, F, G, H M337B, P337H T337B, C, D, E, F, G, H, H-SP | Replacement of installed Slaved Gyro system by ASPEN EFD 1000 PRO Primary Flight Display | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CS-23, Amendment 2 The requirements for environmental protection and the associated cert (...) | Airplane Flight Manual Supplement 2011_028_C337_506, Rev. 0 or later revisions of the above listed documents approved by EASA-Installation Manual 2011_028_C337_500, Rev. 0-Instructions for Continued Airworthiness 2011_028_C337_507, Rev. 1-Master Document List 2011_028_C337_Form 2, Rev. 0 | None | 22.12.2011 |
| 10038951 | AIR ALLIANCE GmbH | | | | 208, 208B | Installation of Honeywell KFD840 Primary Flight Display. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CS-23 Amendment 02 (not required by 21A.101) The requirements for environmental protect (...) | EASA approved Aircraft Flight Manual Supplement (AFMS): 2011_026_C208_0506 Rev. 2 or later revisions of the above listed documents approved by EASA-Installation Manual (IM) with drawings: 2011_026_C208_0500 Rev. 0-Instructions for Continued Airworthiness (...) | None. | 26.03.2012 |
| 10042509 | AIR ALLIANCE GmbH | | | | PA-44-180 PA-44-180T | Replacement of installed Gyro system by ASPEN EFD 1000 PRO Primary Flight Display. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CS-23, Amendment 3 The requirements for environmental protection and the associated cert (...) | Airplane Flight Manual Supplement 2012_014_PA44_506, Rev. 0 or later revisions of the above listed documents approved by EASA - Installation Manual 2012_014_PA44_500, Rev. 0- Instructions for Continued Airworthiness 2012_014_PA44_507, Rev. 0- Master Document List 2012_014 PA44_Form 2, Rev. 0 | None. | 05.12.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|------------------|----------------|----------------------|---------------------------|--|--|--|--|---------------|
| 10043081 | AIR ALLIANCE GmbH | EASA.A.S.03853 | | | PA-31T, PA-31T1, PA-31T2 | Installation of Avidyne EX-500, TWX 670 and MLX 770 acc. AirAlliance MDL2009_003_A8EA in latest approved Revision | | According to AirAlliance Master Document List Doc. No. 2009_003_A8EA Rev. 0, or later approved Revisions and AirAlliance Engineering Order2009_003_A8EA-0500, Rev. 0, or later Revisions To be operated in accordance with AirAlliance AFM-S Documents No.20 (...) | none | 20.05.2009 |
| 10046660 | AIR ALLIANCE GmbH | | | | LEARJET 35A | Replacement of mechanical HSI by Sandel SN4500 EHSI | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Air Alliance, Master Document List: 2013_004_LJ35_Form2, revision 03, dated 27.09.2013 Air Alliance, AFM Supplement: 2013_004_LJ35_0506, revision 02, dated 02.07.2013 Air Alliance, Instructions for Continued Airworthiness: 2013_004_LJ35_0507, revision 00 (...) | This approval is limited for MSN 35A-346 only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 09.10.2013 |
| 10014268 | AIR ALPHA A/S | EASA.A.S.01497 | REV 1 | | BAE HS 748 2A & 2B | Installation of a Honeywell EGPRS MK6 system. | | Shadin Fuel/Airdata Computer Installation Manual PN IM2830 Rev. K. Honeywell TRA-45A Installation Manual 34-40-01 Rev. I. Honeywell MK VI Installation Design Guide Dwg. No. 060-4314-150. Ground test for the MK6 EGPRS Dwg No. 060-4167-158 Rev. A. Honeywell (...) | 1. Existing GPS antenna shall be used with coach splitter. | 15.08.2005 |
| 10030456 | AIR ALPHA A/S | | | | B200, B200GT | Air Alpha STC Definition Document EO-B200-25-145. Flight Inspection System Interface Installation King Air 200 Series. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Air Alpha Master Data List No. MDL EO-B200-25-145, Rev. 06 or later revisions of the above listed documents approved by EASA (...) | None. | 16.06.2010 |
| 10030470 | AIR ALPHA A/S | P-EASA.A.S.03969 | | | 200, B200 AND B200GT | Provision for Flight Inspection use of MLS navigation input | The Certification Basis for the original product remains applicable to this certificate/ approval. CS-23 amend 1 eff. 12 Feb 2009 except FAR Part 23.561 Amdt. 23-7, in lieu of CS23.561. The requirements for environmental protection and the associated certification (...) | Air Alpha A/S Master Data List Numbered: MDL EO-B200-25-129 Rev. 04 dated 04 June 2010. Installation in accordance with the Accomplishment Instructions Ref. ALEO-B200-25-129 Rev 01 dated 19 May 2010. Operation in accordance with Air Alpha A/S Flight Manu (...) | MINIMUM FLIGHT CREW: FIS MLS Navigation may only be used with two (2) flight crew members seated in the cockpit. FIS NOT FOR PRIMARY NAVIGATION. FLIGHT INSPECTION GUIDANCE ONLY | 17.06.2010 |
| 10030471 | AIR ALPHA A/S | P-EASA.A.S.03942 | | | B200 AND B200GT | MD-124F Direction Finder Antenna Installation | The Certification Basis for the original product remains applicable to this certificate/ approval. CS-23 amendment 1 eff. 12 Feb 2009. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the ori (...) | Air Alpha A/S Master Data List MDL EO-B200-25-125 Rev. 03 dated 09 June 2010. Installation in accordance with Accomplishment Instructions Ref. ALEO-B200-25-125 Rev 00 dated 08 July 2009. Instructions for Continued Airworthiness i.a.w. Air Alpha A/S Mainten (...) | None | 17.06.2010 |
| 10014578 | AIR ALPHA MAINTENANCE A/S | EASA.A.S.02683 | | | BEECH B200 | Installation of Trane & Trane Aero-I satcom equipment in accordance with danish STC 03/99. Note: This STC supersedes SLV STC No. 3/99 | | Copenhagen Avionics T.O package: "T & T Aero-I Satcom installation". AFM Supplement: Trane & Trane Aero-I Satcom system in Beech model B200 Danish STC 03/99. | | 13.07.2006 |
| 10014579 | AIR ALPHA MAINTENANCE A/S | EASA.A.S.02684 | | | BEECH B200 | Installation of Trane & Trane Aero-M satcom equipment in accordance with danish STC 04/99. Note: This STC supersedes SLV STC No. 04/99 | | Copenhagen Avionics T.O package: "T & T Aero-M Satcom installation". AFM Supplement: Trane & Trane Aero-M Satcom system in Beech model B200 Danish STC 04/99. | | 13.07.2006 |
| 10014580 | AIR ALPHA MAINTENANCE A/S | EASA.A.S.02685 | | | BEECH B200 | Installation of Trane & Trane Aero-C satcom equipment in accordance with danish STC 05/99. Note: This STC supersedes SLV STC No. 05/99 | | Copenhagen Avionics T.O package: "T & T Aero-C Satcom installation". AFM Supplement: Trane & Trane Aero-C Satcom system in Beech model B200 Danish STC 05/99. | | 13.07.2006 |
| 10014581 | AIR ALPHA MAINTENANCE A/S | EASA.A.S.02686 | | | BEECH B200 | Installation of Trane & Trane Aero-I, Aero-M and Aero-C satcom equipment in accordance with danish STC 29/99. Note: This STC supersedes SLV STC No. 29/99. | | Copenhagen Avionics T.O package: "T & T Aero-I, Aero-M and Aero-C Satcom installation". AFM Supplement: Trane & Trane Aero-I, Aero-M and Aero-C Satcom systems in Beech model B200 Danish STC 29/99. | | 13.07.2006 |
| 10014884 | AIR ALPHA MAINTENANCE A/S | EASA.A.S.03112 | | | CESSNA 560 XL | Installation of UNIFIS 3000 Flight Inspection System: Console, cockpit indication system, antennae | | AFM Supplement FMS.EO-C560XL-25-117 Certification Plan CP-C560XL-25-117. Compliance Checklist CCL-C560XL-25-117. | | 12.12.2007 |
| 10014894 | AIR ALPHA MAINTENANCE A/S | EASA.A.S.03124 | | | BEECH 200, SUPER KING AIR | Replacement of ATC transponders with Rockwell Collins TDR-94 or TDR-94D Mode S systems for ELS and EHS. Installation of top Antenna(s) C1110-660 or DMN 150-6. | | Air Alpha EO-B200-34-104 AFM Supplement: FMS.EO-200-34-104 R1 | | 23.07.2007 |
| 10015164 | AIR ALPHA MAINTENANCE A/S | EASA.A.S.03535 | | | CESSNA 560 XL | Printer Stand Installation | | STC Definition Document EO-B200-C560-25-124 EO-124 Supplemental Maintenance Manual EO-124 Drawing List | | 28.04.2008 |
| 10015217 | AIR ALPHA MAINTENANCE A/S | EASA.A.S.03632 | | | BEECH B200 | Flight Inspection System antenna and equipment interface installation. | | - Installation per Air Alpha Accomplishment Instruction ALEO-B200-25-122 Rev 0 dated 15 October 2008 and Master Drawing List # Antenna & Equipment Interface Installation EO-B200-25-122 dated 6 Nov 2008. Operations per Air Alpha Flight Manual (...) | 1. The applicability of this STC is limited to B200 S/N:BB-1399.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will (...) | 17.04.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|----------------|----------------|-----------------------|---|--|--|---|--|---------------|
| 10017124 | AIR ALPHA MAINTENANCE A/S | EASA.R.S.01211 | | DANISH STC NO.51/2002 | EUROCOPTER FRANCE EC 120 B | Installation of Flying Pictures Model BSS 400 Camera System in accordance with Danish STC No. 51/2002 | | FT Engineering Drawings No.: 500-00-001 dated 1 April 2004 Version 3.FT Engineering HFM Supplement ST 51/2002 dated April 2004 as SLV approved 16 April 2004 | 1. This STC does not include the Videolink System with its Antenna, Antenna Actuator, Power Supply, Remote Control, and Amplifier. Concerning the transfer-CAA-Denmark confirms that the new STC holder, Air Alpha Maintenance A/S, has received all necessary data. CAA-Denmark confirms that Air Alpha Maint | 04.01.2006 |
| 10043035 | AIR ALPHA MAINTENANCE A/S | EASA.R.S.01493 | | | EC-135 | GARMIN 530 COM/NAV/GPS Installation: Optional installation of a GARMINGS530 unit to an existing GNS430 unit and replacing Bendix/King KY196BVHF COM unit. The installation is in accordance with EO-EC135-34-131 Rev. 0 or later approved revision. | | - EC135 Flight Manual Supplements: EC135 Flight Manual Supplement, FMS.EU-EC135-34-131 Rev. 4, dated 06.05.2009, or later approved revision- Technical Instructions: Installation and Maintenance Manual, SMMATP34.0120, Rev. 0 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The (...) | 08.06.2008 |
| 10014960 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.A.S.03231 | | | BEECH 65-A90 BEECH C90 | Installation of Air Medical Interior "Quick Conversion" Beech 65-A90, Beech C90, P/N 65-A90-25-20-15.000. | | # AFM Beech 90 Supplement, Air Ambulance Equipment Beech90 P/N 65-A90-25-20-15.000, AFM AAT 46.# Component-, Installation-, Operation Manual, illustrated Part Catalog Beech90 Air Ambulance Equipment 90 P/N 90-25-20-15.000, CMM-Beech-AAT63 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 19.12.2007 |
| 10015192 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.A.S.03576 | | | BEECH 100 BEECH 200 | Installation of Air Medical Interior. | | Component-, Installation-, Operation-Manual, illustrated part Catalog Beech 100/200 Air Ambulance Equipment, P/N 100/200-25-20-15.000, CMM-Beech-AAT67 Airplane Flight Manual: Beech 100 Supplement, Air Ambulance Equipment Beech 100 King Air/P/N 100-25 (...) | | 17.06.2008 |
| 10017086 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.00461 | REV. 3 | | AS 332 C, AS 332 C1, AS332 L1, AS 332 L2 SA 330 J, EC225 LP | Cabin Interior "Quick Conversion" kits, installing either AAT Air Ambulance Equipment or AAT Mission Equipment cabin layout configurations. Note: This STC is a superseding revision of EASA.R.S.00461 Rev. 2, underlining Associated Technical Documentation updates. | The Certification Basis for the original product remains applicable to this certificate/ approval. | See attached STC Continuation Sheet for Associated Technical Documentation according to cabin conversion kits Parts Number (P/N) and aircraft models | None | 03.08.2010 |
| 10017087 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.00462 | REV. 3 | | EC 135 | Installation of Air Ambulance Equipment, Mission Equipment, Airborne Radiation Measuring Equipment, Revision 2 adds VIP Interior P/N 135-25-20-5000-900 incl. flooring, seats, fairing and cabinets. | | AR 10/97- Component-, Installation-, Operation- Manual, illustrated Part Catalog EC 135 Air Ambulance Equipment P/N 135-25-20-5000, CMM-EC135-AAT11, Supplement to CMM-EC135-AAT11, CMM-EC135-AAT13, CMM-EC135-AAT29-Component-, Installation-, Op (...) | 1. Subject to compliance with the provisions of EC 1702/2003 PART 21 Subpart E, this Certificate and associated data shall remain valid until surrendered, withdrawn or other terminated.2. This STC is approved only for the product configuration (...) | 08.10.2008 |
| 10017112 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.01089 | | | BELL 407 | Mission Equipment Bell 407 # Several racks to install a camera equipment for TV productions | | 1. Certification Base for model 407 FAR part 27, dated October 2, 1964 Amendment 27-1 through 27-30;2. Flight Manual Supplement FMS AAT40 to Rotorcraft Flight Manual BHT-407-FM-13. AAT test report No. 092/05 (Compliance, Analysis and Test, (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 02.11.2005 |
| 10017225 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.01393 | | | AS 332 L2 | VIP interior, quick conversion P/N 332-25-24-3000-901. | | # Definition: Drawings list ref. 332-25-24-3000-901 dated 02/08/2007 as appended in report n°101/07 or later approved revision.# Installation and Maintenance: document CMM-AS332-AAT62, rev.0 dated August 2007 or later revision.# Operation: (...) | 1. STC applicable only on individual AS332L2 aircraft equipped with approved VIP salon consisting of Eurocopter furniture: Forward Lounge: 4 face-to-face VIP seats with 2 VIP tables in the middle based on Eurocopter Dwg 332/851329.Rear Lounge: 8 c (...) | 17.08.2007 |
| 10017254 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.01457 | | | W-3AS | Installation of Air Ambulance Equipment "quick conversion P/N W-3A-25-20-33.000. | | Flight Manual Supplement AAT 47, Revision 0, dated March 2008 Component-, installation-, Operation-Manual, CMM-PZL W-3A-AAT66, Revision 0, dated March 2008. | | 17.04.2008 |
| 10017291 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.01548 | | | AB 139, AW 139 | Installation of Air Ambulance Equipment, Mission Equipment "Quick - Conversion", P/N AW 139-25-20-37.000-60x model | | 1) Rotorcraft Flight Manual Supplement AB/AW 139 Air Ambulance Equipment "Quick Conversion" P/N AW 139-25-20-37.000-60x model;2) Component-, Installation-, Operation-Manual and Illustrated Part Catalog AB/AW 139 Air Ambulance Equipment, P/N AW 139-25-20-37.000-60x model; | 1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2) Th (...) | 09.06.2009 |
| 10031521 | AIR AMBULANCE TECHNOLOGY GmbH | | REV. 1 | | ASTRA SPX | Installation of Air Ambulance Equipment In Astra SPX | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 25 Amdt. 4 (excluding 25.562/25.785 b) (see TGM/25/12 stretchers)The requirements for (...) | For S/N 133: Project Certification Plan /Test report Astra SPX G100-25-20-40.000-5XX, Doc Nr 112/10, Rev 0, dated 08/2010 Component-, Installation-, Operation-, Manual, illustrated Part Catalogue, P/N G100-25-20-40.000-5xx, Rev 0, dated 08/2010 AFM (...) | For S/N 133: Limited to Astra SPX, S/N 133 with Duncan Interior STC ST08088 Will installed, and modified according this EASA STC. This STC requires implementation of Two (2) Minor changes: Avionics Integration & Engineering Corporation AG, CH : Removal of (...) | 13.02.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|----------------|----------------|----------------------|--|---|--|--|---|---------------|
| 10032292 | AIR AMBULANCE TECHNOLOGY GmbH | | | | A330-200/-300 | Air Ambulance Equipment "quick conversion" P/N A330-25-20-10.000 515 - Intensive Care Module P/NA330-25-20-10.300-603 | CS-25 Amdt. 4 except 25.562 addressed per TGM/25/12 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions I (...) | - Compliance Record, ref. doc. CR-A330-10.000-515, Revision 6, dated 25th October 2010 - Component, Installation and Operation Manual, ref. doc. CMM-A330-AAT73, Rev. 1, dated August 2010 - Aircraft Flight Manual Supplement, ref. doc. AFM-AAT 54, Rev. 0r (...) | This STC does not constitute an approval for installation. Compliance with the following requirements will be demonstrated and covered by the DOA responsible for the medical stretcher installation on the aircraft A330 through a design approval. § 25.561 (...) | 25.10.2010 |
| 10034706 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.00470 | REV.1 | | AS 350 B, AS 350 B1, AS 350 B2 AS 350 B3, AS 350 BA, EC 130 B4 | Installation of Air Ambulance Equipment (removable stretcher) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. JAR 27 first issue applicable to the modification (for AS350 models) CS 27 amdt. 2 appli (...) | For AS350 models: Flight Manual AS350 Supplement, Air Ambulance Equipment P/N350-25-24-1000, FMS SUP.AAT3. Appendix 1 to FMS SUP.AAT3, Appendix 2 to FMS SUP.AAT3 or EC130B4 model: Flight Manual EC130 B4 Supplement - Air Ambulance Equipment "quick-convers (...) | none | 20.04.2011 |
| 10037201 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.00464 | REV. 3 | | 1. BHT 212/412/412 EP 2. AB 212/AB 412/412 EP | Installation of Air Ambulance Equipment Extension of existing EASA.R.S.00464 Rev. 2 with Helicopter Model 412EP | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-29, Amdt. ED 2008/10/RM, 17.11.2008 The requirements for environmental protection an (...) | Flight Manual Supplement Rev. 4, FS AAT 14, date: 06/2011 - Instructions for Continued Airworthiness - ICA, CMM-205/212/412-AAT 18, date: 01/2010 or later revisions of the above listed documents approved by EASA - Master Data List Rev. 0, MDL-Bell 412-1, (...) | None | 11.11.2011 |
| 10037231 | AIR AMBULANCE TECHNOLOGY GmbH | | | | MYSTERE FALCON 50 | Installation of Air Ambulance Equipment "quick conversion" in MF50 in the RH Cabin Section of Aircraft. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 25 Amdt 4 ED 2007/020/R, 27.12.2007 The requirements for environmental protection an (...) | Master Data List Rev.1, MDL-MF50-17.000-50x, Date: 20.10.2011 Certification Program Rev.1, CP-MF50-25-20-17.000-50x, Date: 20.10.2011 Classification to Change to Type design Rev.0, CL No.163, Date: 05.08.2011 Compliance Record Rev.0, CR-MF50-17.000-501, (...) | None. | 11.11.2011 |
| 10038064 | AIR AMBULANCE TECHNOLOGY GmbH | | | | LEARJET 45 | Installation of Air Ambulance Equipment | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI D-01, ESF Lear 45 Ambulance Version, Obstruction due to Stretcher Installation to t (...) | Master Data List MDL-LJ45-45.000-501, Rev. 5 - Flight Manual Supplement AFM AAT 59, Rev. 1 - Component, Operation, Installation Manual IPC CMM-LJ45-AAT78, Rev. 1 or later revisions of the above listed documents approved by EASA | Minor Change LR45-19010-050-LP, Installation of EMS power source must be installed prior this modification | 16.01.2012 |
| 10042957 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.A.S.00463 | REV. 1 | | DORNIER 328-100, 328-300 | Installation of Air Ambulance Equipment, VIP interior | | AR 111991 Component-, Installation-, Operation- Manual, illustrated Part Catalog Air Ambulance Equipment "quick conversion" P/N 10.000 Dornier 328 Non Intensive Care Station RH and LH/S CMM-D0328-AAT6 Non Intensive Care Station LH CMM-D0328-AAT9 Intensi (...) | 1. Subject to compliance with the provisions of EC 1702/2003 PART 21 Subpart E, this Certificate and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 2. This STC is approved only for the product configuration as defined i (...) | 21.04.2005 |
| 10042958 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.A.S.00466 | REV. 1 | | JETSTREAM 31132 | Installation of Air Ambulance Equipment | | AR 15/011. Component-, Installation-, Operation- Manual, illustrated Part Catalog Air Ambulance Equipment JETSTREAM 31/32 "quick conversion" P/N J31-25-24-23.000 CMM-A31-AAT332. Airplane Flight Manual JETSTREAM 31/32 Supplemental Air Ambulance Equipment PINJ31-25-24-23.000, AFM-AAT 23 | 1. Compatibility of this installation with previous installed equipment must be determined by the installer. 2. Subject to compliance with the provisions of EC 1702/2003 PART 21 Subpart E, this Certificate and associated data shall remain valid until su (...) | 21.04.2005 |
| 10042959 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.A.S.00731 | | | LEARJET 55/60 | Installation of Air Ambulance Equipment | | AR 17/001. Component, installation, operation manual, illustrated part catalog Air Ambulance Equipment Learjet 60 "quick conversion" P/N LJ60-25-20-20.000, CMM-L60-AAT28, CMM-LJ60-AAT29.2, Airplane Flight Manual Learjet 60 Supplement, Air Ambulance Equipment P/N LJ60-25-20-20.000, AFM-AT17 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate sha (...) | 17.05.2005 |
| 10042960 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.A.S.00733 | | | CHALLENGER 600 | | | AR 16/001. Component, installation, operation manual, illustrated part catalog Air Ambulance Equipment Challenger 60 "quick conversion" P/N CL-25-20-21.000, CMM-L60-AAT30, CMM-AAT31.2, Airplane Flight Manual Challenger Supplement Air Ambulance "quickc (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate sha (...) | 31.05.2005 |
| 10042961 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.A.S.00734 | | | CITATION 5501560 | Installation of Air Ambulance Equipment | | AR 12/991. Component, Installation, Operation Manual, illustrated Part Catalog Air Ambulance Equipment CITATION BRAVO "quick conversion" P/N550-25-24-19.000, CMM-CIT-AAT36 CITATIONV and ultra "quick conversion" P/N 550-25-24-16.000, CMM-CIT-AAT172. Ai (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configuration shall be determined by the installer. This certificate sha (...) | 31.05.2005 |
| 10042962 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.A.S.00735 | | | BEECH 1900D | | | AR 11/991. Component, Installation, Operation Manual, illustrated Part Catalog Air Ambulance Equipment "quick conversion" P/N 10.000 Beech 1900D Air Ambulance Equipment P/N 1900D-25-20-10.000-601, CMM-BEECH-AAT432. Airplane Flight Manual Beech 1900D S (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configuration shall be determined by the installer. This certificate sha (...) | 31.05.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|----------------------|---|--|---------------------|---|--|---------------|
| 10042963 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.00465 | REV. 2 | | 1. BHT 205 2. AB 205A1 | Installation of Air Ambulance Equipment | | AR04/951. Component-, Installation-, Operation- Manual, illustrated PartCatalogBell 205, AB 205 Air Ambulance EquipmentPIN 205-25-20-1000'One Patient Configuration' CMM-BELL205-AA T3'Two Patient Configuration' CMM-BELL205-AAT82. Flight Manual Bell (...) | Prior to installation of this modification the installer must determinethat the interrelationshipbetween this modification and any other previously installedmodification will introduce noadverse effect upon the airworthiness of the product. The inst (...) | 03.08.2007 |
| 10042964 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.00467 | REV. 1 | | EC 155 AND AS 365 N | Installation of Air Ambulance Equipment, VIP Interior, Mission Equipment | | AR 05/961. Component-, Installation-, Operation- Manual, illustrated PartCatalogAir Ambulance Equipment PIN 365-25-24-2000-500 CMM-AS365-AAT4Air Ambulance Equipment PIN 365-25-24-2000-511 CMM-AS365-AAT2 inclSUPPL1Mission Equipment PIN 155-25-24-200 (...) | 1. Compatibility of this installation with previous installed equipmentmust be determined by the installer.2. Subject tocompliance with the provisions of EC 1702/2003 PART 21Subpart E, thisCertificate and associated data shall remain validuntil surre (...) | 21.04.2005 |
| 10042965 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.00468 | REV. 3 | | BK 117 / EC 145 | Installation of Air Ambulance Equipment, VIP Interior | | AR 13/011. Component-, Installation-, Operation- Manual, illustrated PartCatalog BK 117 series AirAmbulance EquipmentPIN 117-25-20-6000-600 CMM-BK117C-2-AAT33PIN 117-25-20-6000-500 CMM-BK117-AAT42PIN 117-25-20-6000-700 CMM-BK117-AAT55Corporate Inter (...) | Prior to installation ofthis modification the installer must determine that interrelationshipbetween this modification and any other previously installedmodification will introduce noadverse effect upon the airworthiness of the product.The instal (...) | 10.12.2007 |
| 10042966 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.00469 | REV. 1 | | AS 355 | Installation of Air Ambulance Equipment | | AR 11/971. Component-, Installation-, Operation- Manual, illustrated PartCatalog AS 355 Air Ambulance Equipment PIN 355-25-24-1000, CMM-AS-AAT0issue 0411995, IM-AS355-AAT01, CMM-AS355-AAT25, CMM-AS355-AAT412. Flight manual AS 355 Supplement, Air Am (...) | 1. Compatibility of this installation with previous installed equipmentmust be determined by the installer.2. Subject tocompliance with the provisions of EC 1702/2003P ART 21 Subpart E, thisCertificate and associated data shall remain validuntil surre (...) | 21.04.2005 |
| 10042967 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.00541 | | | MD 900 SERIES | Installation of Air Ambulance Equipment. | | AR 019/03.1. Component-, Installation-, Operation- Manual, illustrated PartCatalog MD 900 AirAmbulance Equipment PIN 900-25-24-32.00 CMM-MD900-AAT47.2. Flight Manual MD 900 Supplement, Air Ambulance Equipment PIN900-25-24-32.00 FMSAAT35. | 1. This STC is approved only for the product configuration as defined inthe approved designdata referred to in the paragraph "Description". Compatibility withother aircraft/engineconfigurations shall be determined by the installer.This certificate s (...) | 20.01.2005 |
| 10042968 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.00730 | | | S76A | Installation of Air Ambulance Equipment | | AR 14/011. Operation-Manual, Installation, Operation Manual, illustrated partCatalog Air AmbulanceEquipment SIKORSKY S 76A "quick conversion" P/N 76-25-20-34.00.CMM S76-AAT342. Flight manual supplement, Air Ambulance Equipment P/N75-25-20-34.00, FMS-AAT22 | This STC is approved only for the product configuration as defined inthe approved design data referred to in the paragraph "Description".Compatibility with other aircraft/engine configurations shall bedetermined by the installer.This certificate sha (...) | 20.05.2005 |
| 10042969 | AIR AMBULANCE TECHNOLOGY GmbH | EASA.R.S.00732 | | | A109 | Installation of Air Ambulance Equipment | | AR 16/021. Component-<->-Installation-<->-Operation-Manual, illustratedPart Catalog A 109 Series CMM-A-AAT41 Air Ambulance Equipment PIN109-25-20-35.0002. Flight Manual Supplement, Air Ambulance Equipment PIN109-25-20-35.000 FMS AAT 31 | 1. This STC is approved only for the productconfiglrations as defined inthe approved designdata referred to in the paragraph "Description". Compatibility withother aircraft/engineconfigurations shall be determined by the installer.This certificate sh (...) | 26.07.2005 |
| 10043115 | AIR BERLIN PLC & Co. | EASA.A.S.02769 | | | APPLICABLE TO 737 MODELS AS PE DOCUMENT 737-DMD34-01-2006 | 737 -DMD34-0 1-2006: Activation of Enhanced Surveillance Parameter. | | Design Modification Document 737-DMD34-01-2006 Rev IR dtd. 27.11.06 orlater approvedrevision-AFM supplement 737-AFM34-01-2006 Rev. IR dtd. 27.11.2006 or laterapproved revision | 1. Compliance to NP A 20-12a was shown.2. Applicable to 737/models as per Design Modification Document737-DMD34-0 1-2006.3. This STC is approved only for the product configurations as defined inthe approved designdata referred to in the paragraphs "Des (...) | 07.02.2007 |
| 10014296 | AIR CARAIBES INDUSTRIES | EASA.A.S.01540 | | | DORNIER DO 228-202K | Flight Data Recorder installation. Installation of SSFDR type flight data recorder following OPS 1.725 implementation. | | Pilot's Operating Handbook Dornier 228 MSN 8139 Reg. F-OGOL Supplement to MOD Air Carabes Industries DO228-0703, Edition 2 dated 6/04/2005. | 1. This STC is limited to MSN 8139, F-OGOL | 12.09.2005 |
| 10016784 | AIR COMM CORPORATION | EASA.IM.R.S.01125 | | SR00491DE | EUROCOPTER FRANCE EC 120 B | Installation of Cabin Air Conditioning system.The present STC is the validation of the FAA STC n°SR00491DE. | | Definition: Air Comm Corporation Master Drawing List report"DL-EC120, revision D, dated December 26, 2002 approved byFAA or later approved revisionsInstallation: Air Comm Corporation Master Drawing List report n°DL-EC120, revision D, dated December (...) | | 24.01.2006 |
| 10016787 | AIR COMM CORPORATION | EASA.IM.R.S.01132 | | SR00463DE | AGUSTA A-119 | Air Conditioning System for Agusta A 119FAA STC SR00463DE issued on 11th January 2002 and amended on 24th January 2006 | | - Flight Manual Supplement Doc. N° A119-1 rev. 3 dated 14thJuly 2005 or later approved revision.- Instruction for Continued Airworthiness Doc. N° A119-206M-1 rev. 1 dated 29th August 2005 or later approved revision.- Master Drawing List Doc. N° DL-A (...) | This approval should not be extended to rotocraft of this model on which other previously approved modifications are incorporated unless it is determined that the interrelationship between this change and any of those other previously approved mod (...) | 02.03.2006 |
| 10016812 | AIR COMM CORPORATION | EASA.IM.R.S.01173 | | FAA STC SR00222DE | BELL 407 | Installation of a vapour cycle air conditioning system | | Master drawing ListDL-407EC rev ZFlight Manual Supplement 407EC-1 rev NC | None | 30.06.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|---------------------|----------------|----------------------|--|--|--|---|---|---------------|
| 10016939 | AIR COMM CORPORATION | EASA.IM.R.S.01327 | | FAA STC SH4057NM | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of bleed air cabin heater and windshield defrost system. Note: STC SH4057NM was originally approved by the UK CAA under Airworthiness Approval note NO. 25242. This approval adds models S76A and S76B to the approval. This STC supersedes UK CAA Airworthiness Approval note NO. 25242. | | DL-S76H revision H dated 15 December 1995 or later approved revision # Drawing List S76H-100 dated 16 December 1994 or later approved revision # FMS for Bleed Air Cabin Heat, S-76H-102 dated 31 December 1997 or later approved revision # FMS for S76C w/ (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "#Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.05.2007 |
| 10016959 | AIR COMM CORPORATION | EASA.IM.R.S.01356 | | FAA STC SR00601DE | EUROCOPTER EC 135 | Installation of Cabin Air Conditioning System i.a.w. FAA STC SR00665DE, dated 22/06/2005. | | - Master Drawing List DL-EC135, Rev. 1, dated 06/07/2007- Instructions for continued Airworthiness EC135-200M-1, Revision 5, dated 12/07/2007- Flight Manual Supplement EC135-1, Revision 3, FAA approved 16/07/2007 | | 09.10.2007 |
| 10017044 | AIR COMM CORPORATION | EASA.IM.R.S.01468 | | FAA STC SH5226NM | BELL 230 | Installation of a bleed air cabin heating system. | | 230H-100M-7 revision 2 dated 28 October 1993, or later approved revision # Flight Manual Supplement; DL-222H revision J dated 26 August 1991, or later approved revision # Master Drawing List-222H-200M revision B dated 2 August 1993, or later approve (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 02.09.2008 |
| 10026907 | AIR COMM CORPORATION | P-EASA.IM.R.S.01559 | | FAA STC SR00230DE | BELL 407 | Bell Helicopter Model 407 Cabin Ventilation System. | FAR 27 | DL-407V revision L dated 16 April 2009 or later approved revision -Master Document list; 407V-1 revision 2 dated 19 May 2000 or later approved revision - Flight Manual Supplement; 407V-300M revision 0 dated 11 April 2000 or later approved revision - Instructions for Continued Airworthiness. | None | 21.08.2009 |
| 10028717 | AIR COMM CORPORATION | | | FAA STC SR00693DE | BELL HELICOPTER 429 | Installation of Cabin Air Conditioning System Provisions | CS27 Change 1The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | DL-429C, Section 1 revision B dated 5 June 2009 - Master Drawing List-429EC-200M-1 revision 1 dated 9 September 2009 - Instructions for Continued Airworthiness; 429EC-100M revision 0 dated 4 September 2009 - Flight Manual Supplement or later revisions (...) | None | 04.02.2010 |
| 10029555 | AIR COMM CORPORATION | | | FAA STC SR00543DE | EC 130 B4 | Installation of Cabin Air Conditioning System | The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | - Air Comm Corporation Master Drawing List Report No DL-EC-130, Revision U, dated 12/18/2009- Report EC-130-03-10, dated March 16, 2010- Rotorcraft Flight Manual Supplement Document No EC130B4-1 dated April 1, 2004- Rotorcraft Flight Manual Supplement D (...) | None | 12.04.2010 |
| 10030166 | AIR COMM CORPORATION | | | FAA STC SR00463DE | A119, AW119 MKII | FAA STC No. SR00463DE Installation of Air Conditioning System | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | - FM Supplement Doc. No. A119-1, Rev. 6, 11 June 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)-FAA STC No. SR00463DE amended 27 November 2 (...) | None | 31.05.2010 |
| 10038232 | AIR COMM CORPORATION | | | FAA STC SR00692DE | 429 | Installation of Bleed Air Heater | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | 429H-1 issue 1 dated 8 July 2011 - Flight Manual Supplement or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) DL-429H revision F dated 23 June 2009 - (...) | None | 30.01.2012 |
| 10043066 | AIR COMM CORPORATION | EASA.IM.R.S.01549 | | | BELL 412 | Bell Helicopter 412 air Conditioning System. | | - DL-412AC Revision E dated 2 July 1997 or later approved revision -Master Drawing List-412AC-100 dated 10 July 1997 - flight manual supplement is required for Air Comm Corporation drawing 412AC-100 configuration (no temperature control) - 412AC-100 (...) | none | 15.05.2009 |
| 10016847 | AIR CRUISERS COMPANY | EASA.IM.R.S.01223 | | FAA STC SR01842NY | EC 120 B | Installation of Air Cruisers Company Emergency Flotation System. (STC Classified as Basic by the FAA) | | EDR 20-1367 rev C dated 10 December 2003 - Installation Data List EDN 2925 rec IR dated 17 December 2003 - Flight Manual Supplement EDN 2921 rev B dated 8 August 2003 - Instructions for Continued Airworthiness. | This installation is for the Air Cruisers Company Mechanical Activation System to the Eurocopter Emergency Flotation System installed per Eurocopter Service Bulletin Optional No. 25.001, Revision O, dated January 27, 1999 or later approved revision. | 08.03.2007 |
| 10016856 | AIR CRUISERS COMPANY | EASA.IM.R.S.01232 | | FAA STC SR02166NY | AS 350 B, AS 350 B1, AS 350 B2 AS 350 B3, AS 350 BA AS 355 F1, AS 355 F2, AS 355 N | Pneumatic Activation System to the Eurocopter Emergency Flotation System. According to FAA STC n. SR02166NY | | Installation Data List n. EDR 20-1400 rev. E dated February 15, 2006 or later approved revisions/Instructions for continued airworthiness n. EDN 3067 rev. A, dated October 24, 2005 or later FAA approved revisions. Flight Manual Supplement n.# EDN 307 (...) | Applicable to :AS 350 B, B1, B2, BA and B3 a/c equipped with Eurocopter Emergency Flotation System installed per Eurocopter Service Bulletin Optional n. 25.00.35, Rev 3, dated November 23, 1999AS 355 F1, F2 and N a/c equipped with Eurocopter Emergen (...) | 16.03.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|-------------------|----------------|----------------------|---|--|---|--|---|---------------|
| 10017046 | AIR CRUISERS COMPANY | EASA.IM.R.S.01470 | | FAA STC SR02345NY | AS 350 B, AS 350 B1, AS 350 B2 AS 350 B3, AS 350 BA, AS 350 D | Fuselage Mounted Liferaft System (FMLR) installation. | | Installation Data List EDR 01-1438 Rev. B, dated October 30, 2007 or later approved revisions; Flight Manual Supplement EDN 3259, Rev. #4 or later approved revisions; Instructions for Continued Airworthiness ICA-25-62-37 Rev. #4 dated September 18, 2007 or later approved revisions. | Installation of DART Aerospace Spacepod, P/N D350-600-(042,142, 242) side cargo compartment extender law STC EASA.IM.R.S.01292 rev. 1 issued on 07 May, 2008 is required prior to the installation of the Fuselage Mounted Liferaft System; Prior to insta (...) | 18.07.2008 |
| 10029849 | AIR CRUISERS COMPANY | | | FAA STC SR02538NY | BELL 407 | Bell 407 Emergency Float Kit P/N 67432-101 Note: This STC is for the installation of the Pneumatic Activation System only. | The Certification Basis for the original product remains applicable to this certificate/ approval The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | EDN3379 revision IR dated 27 June 2008 - Rotorcraft Flight Manual Supplementor later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) EDR No. 20-1457 revis (...) | For installation only on Bell 407 models previously equipped with the Bell Emergency Float Gear Skid tubes BHT P/N 206-053-184-121 and 206-053-184-122 or approved equivalents. The ACC Bell 407 Emergency Float Installation Kit is used in conjunction with (...) | 26.04.2010 |
| 10029956 | AIR CRUISERS COMPANY | | | FAA STC SR02633NY | BELL 206 L-4 | Bell 206L-4 Emergency Float Kit P/N 67585-101 | The Certification Basis for the original product remains applicable to this certificate/ approval The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | EDN3358 revision IR dated 22 December 2008 - Rotorcraft Flight Manual Supplementor later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) EDR No. 20-1462 (...) | The ACC Bell 206L-4 Emergency Float Installation Kit is for the installation of the Pneumatic Activation System only, and is used in conjunction with the float bags Type Certified per TCCA TC H-92 (Bell P/Ns 206-706-210-113). | 06.05.2010 |
| 10044929 | AIR CRUISERS COMPANY | | | FAA STC SR02254NY | EC130B4 | Installation of Air Cruisers Company (ACC) Emergency Float Installation Kit P/N 65930-101 This STC is the validation of FAA STC SR02254NY (date amended April 30, 2013) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Procedures per Process Specification No P-12314 Revision C, dtd May 17, 2006 ACC Rotorcraft Flight Manual Supplement EDN 3150 Revision 1, dtd 04/30/2013 ACC Instructions For Continued Airworthiness EDN 3131 Revision- original, dtd December 1 (...) | 1 - This STC does not approve the installation of an emergency flotation system. The STC approval is limited to the replacement of the electric means to activate floats of the Emergency Flotation System installed on EC130B4 with the ACC Pneumatic Activ (...) | 21.05.2013 |
| 10014264 | AIR FRANCE | EASA.A.S.01494 | | | B767 | Rockwell Collins Aero I SATCOM installation | | CAS-B767-05-004 Rev 0 | | 19.12.2005 |
| 10014361 | AIR FRANCE | EASA.A.S.01627 | | FAA STC ST02348NY | BOEING 767-300 | Cabin Reconfiguration with an increased seating configuration for LOT Polish Airlines. | | Air France Change Approval Sheet CAS-B767-05-006 Revision 0 or later approved revision. Air France AFM Supplement MA.PM 133/06 KA Initial Release dated 2006-02-17 or later approved revision. | | 24.04.2006 |
| 10014380 | AIR FRANCE | EASA.A.S.01658 | | | BOEING 767 | Emergency Escape Path Marking System | | Air France Industries Modification EA33-B767-05-008 Revision 00 or later approved revision. Air France Industries AFM Supplement reference MA.PM 827/05 CE Initial Release dated December 06 2005 or later approved revision. | | 08.02.2006 |
| 10014601 | AIR FRANCE | EASA.A.S.02724 | | | BOEING 767-300 | Enhanced Surveillance Dual ATC Mode S Installation | | Air France Change Reference CAS-B767-06-007 REV 0 Air France Industries AFM Supplement MT.MO 425/06/ESG REV 0 dated 06 June 2006 or later approved revision | | 23.02.2007 |
| 10017148 | AIR GRISCHA HELIKOPTER AG | EASA.R.S.01251 | REV. 1 | FOCA STC 56-30-01 | AS 350 B3 | Bubble Window RH Installation P/N 030603/350BW according to FOCA Data Sheet 56-30-01. Installation of P/N 030603/350BW according to FOCA data sheet 56-30-01 at issue 3. Revision 1: This is change to EASA.R.S.01251 issued 23/03/2006 which was based on STC (...) | FAA 27. The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | RFMS, bubble window kit 030603/350BW, Eurocopter Helicopter Model AS350B3, issue 2 dated 1/05/2006 or later revisions of the above listed documents approved by EASA-FOCA data sheet 56-30-01 issue 3 dated 10.5.2006- Bubble Window Kit, rev.01 dated 2 (...) | 1. When the bubble window is used for vertical reference, the following applies: If the pilot cannot firmly hold the collective pitch lever when the position intended for vertical reference through the bubble window, an approved collective pitch level (...) | 21.03.2011 |
| 10017167 | AIR MACH S.R.L. | EASA.R.S.01275 | | | EUROCOPTER FRANCE AS 350 B2 | Kit Air Mach ELIRAI TV-99-Internal Rack/ Installation of electronic equipment for radio bridge | | Master Drawing List MDL-INT-001 Rev. 0 Dated 26 April 2006 Instruction for Continues Airworthiness ICA-001 Rev. 0 Dated 26 April 2006 Part List Electronic Equipment PL-TV-99 Rev. 0 Dated 26 April 2006 RFM Supplement S-RFM-001 Rev. 0 Dated 26 April 2006 | This installation should not be incorporated in any aircraft unless it is determined that the interrelationship between this installation and any previously approved configuration will not introduce any adverse effect upon the airworthiness of the aircraft | 10.05.2006 |
| 10017168 | AIR MACH S.R.L. | EASA.R.S.01276 | | | EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 | Kit Air Mach ELIRAI TV-99-Internal Rack/ Installation of electronic equipment for radio bridge | | Master Drawing List MDL-INT-001 Rev. 0 Dated 26 April 2006 Instruction for Continues Airworthiness ICA-001 Rev. 0 Dated 26 April 2006 Part List Electronic Equipment PL-TV-99 Rev. 0 Dated 26 April 2006 RFM Supplement S-RFM-001 Rev. 0 Dated 26 April 2006 | This installation should not be incorporated in any aircraft unless it is determined that the interrelationship between this installation and any previously approved configuration will not introduce any adverse effect upon the airworthiness of the aircraft | 10.05.2006 |
| 10016871 | AIR METHODS CORPORATION | EASA.IM.R.S.01247 | | FAA STC SR00032DE-D | AGUSTA BELL 412 BELL 412 | Installation of Air Methods Corporation Emergency Medical System interior that provides 3 patient locations to be configured one off two ways using an Intensive Care Transport Litter (ICTL) and Ferro stretchers. | | MDL 141-9009 rev B dated 10 June 1993. | 1. Single and dual patient configurations are to be installed in strict accordance with Air Methods Corporation Maintenance Manual AMMS 050-0110 rev NIC dated 156 April 1993 or later approved revision and FAA approved Rotorcraft Flight Manual Sup (...) | 11.01.2007 |
| 10029569 | AIR METHODS CORPORATION | | REV. 1 | FAA STC SR00705DE-D | S-76A, S-76B, S-76C | Installation of Articulated Patient Loading System (APLS). Note: The STC amendment adds an ALLFA litter as an option. | As described in FAA TCDS H1NE The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval 14CFR Part 29 §29.785(a), (b), (e), (k)(1) <-> (k)(2), which (...) | -Rotorcraft Flight Manual Supplement No. 080-0476, revision A, dated June 9, 2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) Master Drawing L (...) | Required pre-modification Sikorsky cabin medical interior GA dwg. no. 09076B10E001-1 | 19.08.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|----------------|----------------|----------------------|---|--|---|---|---|---------------|
| 10030372 | AIR METHODS CORPORATION | | | FAA STC SR00250DE-D | MD900 | Installation of an EMS Interior in a MD902 Helicopter | As described in FAA TCDS H19NMThe Certification Basis for the original product remains applicable to this certificate/ approvalThe certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | -Master Drawing List Report No. 231-9000-001, revision J, dated March 5, 2010 - Rotorcraft Flight Manual Supplement "EMS Interior", No. 080-0209, revision E, dated June 5, 2008 - Rotorcraft Flight Manual Supplement "Aft Equipment Frame", No. 080-0220, rev (...) | If damage or excess wear warrants structural repair or alteration to the floor system, pre-approval authorized by the holder of this STC (required in accordance with AMMS 050-0167, Rev. (), dated June 13, 1996 or later FAA accepted revision. The mult (...) | 11.06.2010 |
| 10031942 | AIR METHODS CORPORATION | | | FAA STC SR00702DE-D | BELL 429 | Installation of Supplemental Strobe Lights | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | 080-0471 revision IR dated 11 March 2010 - Flight Manual Supplement; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) 291-9016 issue IR dated 1 Feb (...) | None | 23.09.2010 |
| 10032603 | AIR METHODS CORPORATION | | | FAA STC SR00701DE-D | 429 | Installation of a controllable Searchlight | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 291-9010 revision IR dated 3 February 2010 - Master Drawing List; 080-0468 revision IR dated 15 March 2010 - Flight Manual Supplement; 050-0537 revision A dated 23 February 2010 - Instructions for Continued Airworthiness or later revisions of the above (...) | none | 17.11.2010 |
| 10032617 | AIR METHODS CORPORATION | | | FAA STC SR00704DE-D | BELL 429 | Installation of Emergency Medical Services Interior and Systems | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | 291-9002 revision IR dated 9 March 2010 - Master Drawing List; 080-0464 revision IR dated 17 March 2010 - Flight Manual Supplement; 050-0530 revision C dated 4 March 2010 - Instructions for Continued Airworthiness or later revisions of the above listed (...) | None | 17.11.2010 |
| 10032620 | AIR METHODS CORPORATION | | | FAA STC SR00710DE-D | BELL 429 | Installation of Scene and Patient Loading Lights | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | 291-9012 revision IR dated 1 February 2010 - Master Drawing List; 080-0469 revision IR dated 17 February 2010 - Flight Manual Supplement; 050-0538 revision A dated 23 February 2010 - Instructions for Continued Airworthiness or later revisions of the above (...) | None | 17.11.2010 |
| 10032621 | AIR METHODS CORPORATION | | | FAA STC SR00698DE-D | BELL 429 | Installation of Patient Loading System | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | 291-9004 revision IR dated 1 March 2010 - Master Drawing List; 080-0465 revision IR dated 12 March 2010 - Flight Manual Supplement; 050-0532 revision A dated 4 March 2010 - Instructions for Continued Airworthiness or later revisions of the above listed (...) | None | 17.11.2010 |
| 10040541 | AIR METHODS CORPORATION | | | FAA STC SR00523DE | SA365N, SA365N1, AS365N2 | Installation of Emergency Medical Interior | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -080-0295 revision A dated 18 May 2012, Flight Manual Supplement or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)-241-9001 revision J dated 18 May (...) | None | 09.07.2012 |
| 10040543 | AIR METHODS CORPORATION | | | FAA STC SR00525DE | SA365N, SA365N1, AS365N2 | Installation of two (2) aft jump seats | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -080-0295 revision A dated 18 May 2012, Flight Manual Supplement or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)-241-9007 revision B dated 2 Sept (...) | None | 09.07.2012 |
| 10040545 | AIR METHODS CORPORATION | | | FAA STC SR00524DE | SA365N, SA365N1, AS365N2 | Installation of Articulating Patient Loading System | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -080-0295 revision A dated 18 May 2012, Flight Manual Supplement or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)-241-9003 revision F dated 18 May (...) | None | 09.07.2012 |
| 10014327 | AIR RADIO SERVICE GMBH | EASA.A.S.01580 | | | CESSNA 560 CITATION S/N 560-0260 AND ON | Upgrade of the existing MK V1 Ground Proximity Warning System to MK VIII (Enhanced Ground Proximity Warning System) Cessna 560 Citation Ultra, S/N 560-0260 and on, equipped with Honeywell Integrated Avionics System, Honeywell Primus © II SRZ Integrated Radio System and MK VI GPWS | | ARS Engineering Order 5426/05/2 | N/A, Demonstration on S/N 0446 | 13.03.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|-------------------|----------------|----------------------|--|--|--|--|---|---------------|
| 10014328 | AIR RADIO SERVICE GMBH | EASA.A.S.01581 | | | CESSNA 560 CITATION S/N 560-0260 AND ON | Integration of a Traffic Alert and Collision Avoidance System and update of the ATC Mode S Transponders to Elementary Mode S Cesna 560 Citation Ultra, S/N 560-0260 and on, equipped with Honeywell Integrated Avionics System, Honeywell Primus © II SRZ Integrated Radio System | | ARS Engineering Order 5426/05/1 | N/A: Demonstration on S/N 0446 | 13.03.2006 |
| 10014728 | AIR RADIO SERVICE GMBH | EASA.A.S.02906 | | | LEARJET 55C | Upgrade of the SSR Transponder System to Elementary (Enhanced) Surveillance) | | Certification Program LX-55-ARS-001, initial issue AFM Supplement L55X-ARS34-001-M1-AFM, initial issue | s/n 55-135 and on | 08.03.2007 |
| 10014813 | AIR RADIO SERVICE GMBH | EASA.A.S.03021 | | | CESSNA 560 S/N 560-0111 | Mode S Elementary/Enhanced Surveillance Upgrade | | Declaration of Compliance C560-ARS34-001-M0-DOC, AFM Supplement C560-ARS34-001-M1-AFM, Compliance Report CR-C560-ARS34-001 | EHS compliance is partial. See AFM Supplement for Limitations. | 27.03.2007 |
| 10014826 | AIR RADIO SERVICE GMBH | EASA.A.S.03039 | | | C650 | Mode S Elementary/Enhanced Surveillance Upgrade. | | # Certification Report C650-ARS34-001.# Instructions for continued airworthiness C650-ARS34-001M1-IFCA.# AFM Supplement C650-ARS34-001-M1-AFM.# Engineering Bulletin EB-C650-ARS34-001. | 1. Serial number : 650-7001 and on2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and #Associated Technical Documentation#. Compatibility with other (...) | 29.05.2007 |
| 10014827 | AIR RADIO SERVICE GMBH | EASA.A.S.03040 | | | C650 | Upgrade of a MkVI GPWS to a MkVII EGPWS | | Certification Report C650-ARS34-002Instructions for Continued Airworthiness C650-ARS34-002M1-IFCAAFM Supplement C650-ARS34-002-M1AFM, Rev.1.Engineering Bulletin EB-C650-ARS-002 | Serial number: 650-7001 and on | 29.05.2007 |
| 10015088 | AIR RADIO SERVICE GMBH | EASA.A.S.03423 | | | CESSNA 525 | Rockwell Collins MDF-124F (V2) Direction-Finder-System Installation | | Certification Report C525-ARS34-004, Rev. 1, 15.02.2009Engineering Bulletin EB-C525-ARS34-004, Rev. 1, 11.02.2008AFM Supplement C525-ARS34-004-M1-AFM, Rev. 1, 07.02.2008Instructions for Continued Airworthiness C525-ARS34-004-M1-IFCA, Rev. 0, 31.01.2008 | Refer to AFMS limitation | 27.02.2008 |
| 10015180 | AIR RADIO SERVICE GMBH | EASA.A.S.03558 | | | 550 | Upgrade of SSR Transponder System to Enhanced Surveillance. | | Certification Report C550-ARS34-003, Rev. 1, 23.06.2008Engineering Bulletin EB-C550-ARS34-003, Rev. 0, 02.05.2008Airplane Flight Manual Supplement C550-ARS34-003-M1-AFM, Rev. 1, 23.06.2008Instruction for Continued Airworthiness C550-ARS34-003-M1-IFCA, Rev. 0, 02.05.2008 | | 26.06.2008 |
| 10015209 | AIR RADIO SERVICE GMBH | EASA.A.S.03624 | | | 550 | Installation of an L-3, Model F1000 Solid State Flight DataRecorder including accessories.Amendment of Austrian STC ACG.21E.062. | | ∫ Compliance Report CR-C550-ARS31-001, Rev. 0, 04.06.2008∫ Engineering Bulletin EB-C550-ARS31-001, Rev. 0, 20.05.2008∫ Maintenance Manual Supplement C550-ARS31-001-M1-MMS, Rev. 0, 21.05.2008∫ Airplane Flight Manual Supplement C550-ARS31-001-M1-AFM, Rev. 0, 04.06.2008 | None. | 30.06.2008 |
| 10042947 | AIR RADIO SERVICE GMBH | EASA.IM.A.S.01075 | | | CESSNA 550B | Upgrade of GPWS to EGPWS | | - AFM Supplement 5387/04/1-06 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 10.02.2005 |
| 10017134 | AIR SERVICE CENTER S.R.L. | EASA.R.S.01228 | | | A119 | ASC 119-01-00 #Helibasket Kit#/Installation of External Basket | | ∫ Report Helibasket ASC 119-01-00 ED. 1 dated 23 June 2006Part. 1A) Description of modificationPart. 1B) DrawingsPart. 1C) Weight & Balance ∫ Report Helibasket ASC 119-01-00 Part. 3 #Manuale dell'Utente# composed of:Chapter 1) ServiceBulletin (...) | Report Helibasket ASC 119-01-00 Part.3 Chapter 4 RFM Supplement Ed.2 dated 21 August 2006 | 24.10.2006 |
| 10017154 | AIR SERVICE CENTER S.R.L. | EASA.R.S.01260 | | | A119 | ASC 119-02-00 External self moving mirror kit/Installation of external mirror in accordance with "Certification Report" Edition 0 dated 5 February 2007 Rev. 1 dated 19 March 2007. | | # User Manual, Chapter 1 "Installation Bulletin" Ed. 0 dated 05.02.2007 Rev.1 dated 19 March 2007# User Manual, Chapter 2 "Maintenance Program" Ed. 0 dated 05.02.2007# User Manual, Chapter 3 "IPC" Ed. 0 dated 05.02.2007# User Manual, Cha (...) | This installation should not be incorporated in any aircraft unless it is determined that the interrelationship between this installation and any previously approved configuration will not introduce any adverse effect upon the airworthiness of the aircraft. | 30.04.2007 |
| 10034544 | AIR SERVICE CENTER S.R.L. | | REV. 1 | | 182 SERIES | Installation of Vaccine Dispensal Kit. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated-certificated noiseand/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AFM Supplement & Maintenance Manual Supplement for Air Service CenterKit ASC182-P001-00, included in Doc. ASC182-P001-00-Owm-01or later revisions of the above listed documents approved by EASA. | None. | 11.05.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue | |
|--------------|----------------------------------|----------------|----------------|----------------------|---|---|---|---|---|--|------------|
| 10034547 | AIR SERVICE CENTER S.R.L. | | REV. 1 | | AS 350 B3 | Installation of Vaccine Dispersal Kit. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AFM Supplement & Maintenance Manual Supplement for Air Service Center Kit ASC350-P003-00, included in Doc. ASC350-P003-00-Cwm-01 for later revisions of the above listed documents approved by EASA. | None. | 11.05.2011 | |
| 10014844 | AIR SERVICE VAMDRUP APS | EASA.A.S.03062 | REV. 2 | | PC-12/45 PC-12/47 | Oxygen fill point modification (to remove limitation for installation of STC 10014844 (previous EASA.A.S.03062). | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: As per EASA STC 10014844 (former EASA.A.S.03062) The requirements for environmental protection and the associated certificated (...) | Engineering order EO-PC12-31-121, 1 March 2007; Masterdrawing EN-PC12-31-121-01; Airplane Flight Manual supplement, FMS EO-PC12-31-121; Supplemental Maintenance Manual ATP31.0114 Rev.0, 21 June 2007; MEL Supplement EO-PC12-31-121_Report (...) | Only eligible for PC-12/45 and PC-12/47 without the Oxygen system installation in the Rear fuselage, unless modification "Oxygen fillpoint modification (to remove limitation for installation of STC10014844 (previous EASA.A.S.03062)" is installed. The (...) | 03.06.2013 | |
| 10014992 | AIR SUPPORT INTERNATIONAL S.r.l. | EASA.A.S.03288 | | | PIPER PA-31T1 | ASI BMA 04/B/06 #Installation of Garmin GNS 430 & GNS 530, Goodrich WX 500, Garmin MX 20 and Accessories# | | AFM Supplement No. BMA 04B06 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 22.10.2008 | |
| 10015293 | AIR SUPPORT INTERNATIONAL S.r.l. | EASA.A.S.03835 | | | MOONEY M20K | ASI BMA 01/B/09 # Installation of Garmin GNS 430W for B-RNAV Operation# | | ASI BMA 01/B/09AFMS BMA 01/B/09 | 1) The approval holder shall fulfill the obligations of Part 21, Paragraph 21.A1092) Limited to S/N 25-0862. | 10.06.2009 | |
| 10030246 | AIR SUPPORT INTERNATIONAL S.r.l. | | | | PA -30 | ASI BMA 01/B/10 "Installation of Garmin GNS 430W approved for B-RNAV and other avionics equipment". | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The applicant elected to comply with CS-23 par. 1301, 1309, 1321, 1322, 1329, 1351, 1353 (...) | AFM Supplement Garmin GNS 430W Instructions for Continued Airworthiness BMA 01/B/10 for later revisions of the above listed documents approved by EASA | None. | 07.06.2010 | |
| 10030248 | AIR SUPPORT INTERNATIONAL S.r.l. | | | | PA-30 | | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The applicant elected to comply with CS-23 par. 1301, 1309, 1321, 1322, 1329, 1351, 1353 (...) | AFM Supplement Aspen Avionics EFD 1000 Pro Instructions for Continued Airworthiness BMA 04/B/09 for later revisions of the above listed documents approved by EASA | None. | 07.06.2010 | |
| 10017141 | AIR WORK FRANCE S.A.R.L. | EASA.R.S.01243 | | | AS 355 F, AS 355 F1, AS 355 F2 AS 355 N | Installation of Thales AGILE 2 external camera and associated cabin electronic control unit. | | | 1. Definition: AIRWORK Conformity Demonstration File n° AWF.001.05A revision A dated April 2007.2. Installation and continued airworthiness: AIRWORK Instructions for Installation and Continued Airworthiness n°. ICA-AWF-01A, edition 1 dated Apr (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 27.11.2008 |
| 10017142 | AIR WORK FRANCE S.A.R.L. | EASA.R.S.01244 | | | AS 350 B, AS 350 BA AS 350 B1, AS 350 B2 AS 350 B3 | Installation of Thales AGILE 2 external camera and associated cabin electronic control unit, in accordance with AIRWORKSTC file n° AWF.001.05 revision A, dated January 2007, or later EASA approved revision. | | | Installation: AIRWORK Instructions for Installation and Continued Airworthiness ref. ICA-AWF-01, edition 1 dated January 2007, or later revision. Continued Airworthiness: AIRWORK Instructions for Installation and Continued Airworthiness ref. ICA-AWF-0 (...) | 27.02.2007 | |
| 10017190 | AIR WORK FRANCE S.A.R.L. | EASA.R.S.01327 | | | AS 350 B, AS 350 B1 AS 350 B2, AS 350 BA AS 350 B3 | Nouveau système de prises de vues frontales (Front Mounted Camera Installation). | | DDC AWF 002.06 Flight Manual Supplement AFMS-AWF-01 | | 16.06.2008 | |
| 10017191 | AIR WORK FRANCE S.A.R.L. | EASA.R.S.01328 | | | AS 355 E, AS 355 F, AS 355 F1 AS 355 F2, AS 255 N, AS 355 NP | Nouveau System de Prises de Vues Frontales (Front Mounted Camera Installation) | | DDC AWF 002.06 Flight Manual Supplement AFMS # AWF # 02 dated June 2008 or later EASA approved revision | This STC is not approved for Category A operations. | 16.06.2008 | |
| 10014142 | AIRBERLIN TECHNIK GmbH | EASA.A.S.01256 | REV. 1 | | A330-322 | Cabin reconfiguration from a 18 B/C + 331 Y/C layout to a 18 B/C + 343 Y/C layout. This STC installs a new cabin arrangement. The Revision 1 retains the certificated design of the EASA.A.S.01256 and capture the transfer of STC ownership from LTU (owned (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. As per EASA TCDSA.004 Elect to Comply with CS25 (Original Issue) requirements: 25.811 (...) | Master Document List, ref. MDL330-25-01-05 Revision 2, dated 27 Apr. 2005 or later revisions of the above listed documents approved by EASA | none | 22.05.2012 | |
| 10014924 | AIRBERLIN TECHNIK GmbH | EASA.A.S.03172 | REV. 1 | | A330-322 | Business Class extension to a 30 B/C + 295 Y/C layout. This STC modifies cabin arrangement from a 18 B/C + 343 Y/C layout to a 30B/C + 295 Y/C layout with new B/C seat and stowage compartment installations. The Revision 1 retains the certificated design o (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS25 Amdt 3 The requirements for environmental protection and the associated certificate (...) | Master Document List, ref. MDL330-25-01-07 Revision 1, dated 25 Oct. 2007 or later revisions of the above listed documents approved by EASA | none | 22.05.2012 | |
| 10037159 | AIRBERLIN TECHNIK GmbH | | | | A320-214 | A320 Cabin Reconfiguration for AZAL | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. JAR 25 Change 11 and EASA CS 25.853 Initial Issue, as per Air Berlin Technik's Certificat (...) | Air Berlin Technik's - Master Documentation List A320-MDL25-06-2011 Revision Initial dated 05th November 2011, and Engineering Order A320-EQ25-06-2011 Revision 1 dated 21st October 2011 or later revisions of the above listed documents approved by EASA | This design change (STC) is applicable to the following Airbus A320-214 serial number aircraft (MNS): 2685; 2646; 2653; 2665; 2991 and 3006. | 07.11.2011 | |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------|----------------|----------------|----------------------|---|--|---|---|--|---------------|
| 10039755 | AIRBERLIN TECHNIK GmbH | EASA.A.S.00504 | REV. 1 | | A330-322 | Cabin reconfiguration from a 387 Y/C layout to a 18 B/C + 331 Y/Clayout. This STC installs a new cabin arrangement with in-seat video and additional galley installations. The Revision 1 retains the certificated design of the EASA.A.S.00504 and capture t (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. As per EASA TCDSA.004/Elect to Comply with CS25 (Original Issue) requirements :25.811 (...) | Master Document List, ref. MDL330-25-01-04 Revision 1, dated 09 Dec.2004 | None | 22.05.2012 |
| 10041988 | AIRBORNE MAINTENANCE AND | | | FAA STC ST01096CH | 767-200/-300 | Upgrade of Teledyne Digital Flight Data Acquisition Unit | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this approval. | AMES Master Drawing List RPT-99-06040, Revision G, 22/04/2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | none | 30.10.2012 |
| 10041989 | AIRBORNE MAINTENANCE AND | | REV. 1 | FAA STC ST02325CH | 767-200/-300 | Installation of Electronic Standby Instrument System (ESIS) | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this approval. | ABX Air Master Data List RPT-06-07507, Revision B, 16/08/2006 AFM Supplement, ABX Air, Inc. Document No. RPT-06-07516, Revision IR<->-<-> 06/09/2006 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2 (...) | Only to be installed on a/c with EASA STC 10036223 installed. | 05.11.2012 |
| 10033764 | AIRBORNE TECHNOLOGIES GmbH | | REV. 1 | | P.68TC "OBSERVER" | Change of cabin layout /For the installation of different mission equipment in a Vulcanair P.68TC aircraft, the aircraft cabin has to be modified and provisions to accommodate Mission Equipment have to be installed. This modification includes the removal (...) | Elected to comply: CS 23, Amdt. 2 and FAR 23, effective 1st Feb. 1965 or 23.337, 23.341 and 23.561 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the a (...) | Engineering Order ABVP68-EO-2500-01-03, Issue 3 AFM Supplement ABVP68-AM-2500-01-02, Issue 2 Interface Definition ABVP68-DE-2500-02-02, Issue 2 Technical Order ABVP68-TO-2460-01-01, Issue 1 or later revisions of the above listed documents approved by EASA | Limited to S/N 454, 456 and 459 | 04.01.2012 |
| 10033844 | AIRBORNE TECHNOLOGIES GmbH | | REV. 2 | | T182T | Camera Hatch Installation | The Certification Basis for the original product remains applicable to this certificate/ approval. FAA Certification Basis 3A13; FAR 23 effective February 1, 1965, as amended by 23-1 through 23-6 except as otherwise defined in the TCDS. However the applic (...) | 2058039-EO, Issue 2; Engineering Order 2058039-CA, Issue 1; Continuous Airworthiness Instructions 2058039-MM, Issue 1; Maintenance Manual or later revisions of the above listed documents approved by EASA | none | 24.05.2012 |
| 10034631 | AIRBORNE TECHNOLOGIES GmbH | | REV. 2 | | P2006T | Title: STC-AT-003-Rev2, Provisions for surveillance configuration. Description: Installation of a camera hatch, electric doors, 28V mission power system, electric camera lift and a ventral fin in various combinations. For details refer to section 1.0 (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Depending on aircraft instrument panel configuration (analog instruments or Garmin G950) and extent of modification (electric camera lift installed or not) one or more of the following documents are required: Installed in accordance with "Provisions fo (...) | The installation of steps 1, 3, 5, or 7 of this STC must be carried out while production of the aircraft at the facilities of the airplane manufacturer. Conversions from step 1 or 3 to step 5 or 7 of this STC can either be performed while production of t (...) | 29.01.2013 |
| 10036443 | AIRBORNE TECHNOLOGIES GmbH | | | | P.68 "OBSERVER 2" P.68 "OBSERVER" P.68TC "OBSERVER" | Mission cabin configuration /For the installation of different mission equipment in a Vulcanair P.68 aircraft, the aircraft cabin has to be modified and provisions to accommodate Mission Equipment have to be installed. The installation of mission equipment itself is not part of this STC. | Certification basis as per EASA TCDS: FAR 23, applicable Amendment The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Applicant elected to (...) | Installed in accordance with Change Dossier Doc. No. STC-AT-036, Rev. 2 and Installation Instructions Doc. No. 036-II-STC-AT, Rev. 0 Operated in accordance with Flight Manual Supplement AFM-STC-AT-036, Rev. 1 Maintained in accordance with Maintenance Ma (...) | None | 08.09.2011 |
| 10038317 | AIRBORNE TECHNOLOGIES GmbH | | | | P.68 "OBSERVER 2" P.68 "OBSERVER" P.68TC "OBSERVER" | Electric Camera Lift installation/Installation of an electric camera lift, an operator console and a camera lift control panel. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-23, Amendment 2 The requirements for environmental protection and the associated certific (...) | Change Dossier "Electrical Camera Lift" Doc. No. STC-AT-038, Rev. 1-Installation Instructions Doc. No. 038-II-STC-AT, Rev. 0 Flight Manual Supplement AFM-STC-AT-038, Rev. 1-Maintenance Manual Supplement AMM-STC-AT-038, Rev. 1 or later revisions of the above listed documents approved by EASA | Airborne Technologies STC-AT-036 must be installed prior or concurrent with this modification | 06.02.2012 |
| 10038324 | AIRBORNE TECHNOLOGIES GmbH | | | | P2006T | Hatch Cover. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installed in accordance with Supplemental Type Certificate STC-AT-037, Rev. 2 Maintained in accordance with AMM Supplement AMM-STC-AT-037, Rev. 2 or later revisions of the above listed documents approved by EASA. | STC-AT-003 must be installed prior this modification. | 07.02.2012 |
| 10039513 | AIRBORNE TECHNOLOGIES GmbH | | | | P2006T | Police cabin configuration/Installation of a camera lift for 10' gimbals, operator console and replacement of hatch doors by fixed covers For standard analogue instruments equipped aircraft step 1 of this STC is applicable, for Garmin G950 equipped aircraft step 2 is applicable | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Depending on aircraft instrument panel configuration (analog instruments or Garmin G950) the following documents are required: "Police cabin configuration", Doc. No. STC-AT-045, Rev. 2 "Installation Instructions", Doc. No. 045-II-STC-AT, Rev. 0 AFM Su (...) | Airborne Technologies STC no. "STC-AT-003", step 1 (analog instruments) or step 3 (Garmin G950 version) of latest approved revision must be installed prior this modification. | 08.05.2012 |
| 10040198 | AIRBORNE TECHNOLOGIES GmbH | | | | P2006T | Title: Provisions for radar operation Description: Installation of a radome below the photogrammetric hatch, activation of the rear hatch (installed with EO-AT-047), installation of a ventral fin, upgraded mission power supply system and removal of hat (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certificated noise and/ or emissions levels. | Change Dossier, STC-AT-044, Rev. 3 Supplement to the AFM, AFM1-STC-AT-044, Rev. 2 Supplement to the AMM, AMM1-STC-AT-044, Rev. 1 Installation instructions, 044-II-STC-AT, Rev. 0 or later revisions of the above listed documents approved by EASA | STC-AT-003 step 1 or step 3 and EO-AT-047 must be installed prior this modification. | 18.06.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------|----------------|----------------|------------------------|--|--|--|--|--|---------------|
| 10040955 | AIRBORNE TECHNOLOGIES GmbH | | | | P.68 C, P.68 C-TC | Title: Mission cabin configuration P.68 C /Description: This STC consists of 3 steps:Step 1: To install different mission equipment in a Vulcanair P.68aircraft, step 1 modifies the aircraft cabin and installs provisions to accommodate Mission Equipme (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Change Dossier - Supplemental Type Certificate' Doc. No. STC-AT-050 Rev. 3Supplement to the AFM - Step 1, Doc No. AFM1-STC-AT-050, Rev. 1Supplement to the AFM - Step 2, Doc No. AFM2-STC-AT-050, Rev. 0Supplement to the AFM - Step 3, Doc No. AFM3-STC-AT- (...) | Vulcanair SB-163 must be carried out prior this STC. | 07.08.2012 |
| 10042009 | AIRBORNE TECHNOLOGIES GmbH | | | | EC135 P2+ | Title: Provisions for surveillance operationsDescription: Installation of provisions for surveillance operations, consisting of the following parts:Step 1: Provisions for surveillance operationsStep 2 : Cabin provisionsStep 3 : Trim mass installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Elected to comply to CS 27 Amd. 2The requirements for environmental protection and th (...) | Applicable Documents - Supplemental Type Certificate (Change Dossier), Doc. No. STC-AT-054, Rev. 5, dated 29.10.2012; Installation Instructions, Doc. No. 054-I-STC-AT, Rev. 1, dated 29.10.2012; AFM Supplements: Doc. No. AFM1-STC-AT-054, Rev. 1, dated 2 (...) | None. | 30.10.2012 |
| 10013925 | AIRBUS S.A.S. | EASA.A.S.00196 | REV. 1 | | AIRBUS A319-112 | Cabin reconfiguration into an executive 44 pax, in accordance with the Modification Approval Sheet (MAS) Ref. SOG-00-830, issue 4, dated September 20, 2004 | | | The Document Ref.: DIT/N/2004/1726/OS Revision B must be incorporated as a supplement to the Airplane Flight Manual (AFM) | 02.07.2008 |
| 10013936 | AIRBUS S.A.S. | EASA.A.S.00347 | REV. 1 | | AIRBUS A330-203 | VIP Cabin Installation | | All documents related to this STC are listed in the Master Data List Drawing No MDL-01-502 Issue A (or later EASA-approved revision). | 1. The approval of this modification is limited to the Airbus A330-203 MSN 487 which is used as a private aircraft. The limitations and conditions defined in the EASA Datasheet are applicable. A copy of this STC and of the MDL, AFM must be enclosed (...) | 04.07.2008 |
| 10013937 | AIRBUS S.A.S. | EASA.A.S.00348 | REV. 1 | | AIRBUS A330-203 | Installation of a SATCOM Live TV system. | | All documents related to this STC are listed in the Master Data List Drawing No. MDL-01-580 Issue A. (or later EASA-approved revision). | 1. The approval of this modification is limited to the AIRBUS A330-203 MSN 487. The limitations and conditions defined in the EASA Datasheet are applicable. A copy of this STC and of the MDL, AFM must be enclosed in the file of the modified aircr (...) | 04.07.2008 |
| 10013945 | AIRBUS S.A.S. | EASA.A.S.00387 | REV. 5 | LBA STC TA0441 ISSUE 4 | A300B4-601, -603, -605R A300B4-620, -622, -622R | EASA STC EASA.A.S.00387 revision 4 (ref. to former Luftfahrt Bundesamt STC TA0441 Issue 4)The STC EASA.A.S.00387 has covered the transfer of the STC TA0441 Issue4 (original Aviation Authority reference) from the original Aviation Authority (LBA) to th (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approvalSee Certification Summary Document referenced in Associated Technical Documentation sect (...) | Certification Summary Document EAA2/400/000/001/COS Issue08, dated 16 January 2007- Airbus Deutschland Engineering Order Nr. PFM 00400A Issue 08, dated 07 May 2007- Aircraft Flight Manual Supplement for Airbus A300-600, Revision 07 dated 15 January (...) | N/A | 05.03.2012 |
| 10013946 | AIRBUS S.A.S. | EASA.A.S.00403 | REV. 1 | | A310-304 | - EASA STC EASA.A.S.00403 revision 1 (Ref. to former DGAC-F STC S1-72-01)- The STC EASA.A.S.00403 has covered the transfer of the STC S1-72-01 (Original Aviation Authority reference) from the original Aviation Authority (DGAC-F) to the EASA allowing th (...) | - The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- See Determination of Certification Basis Document referenced A/BWT/C 412.0457/95 The (...) | Certification Summary Document, ref: A/BWT/C 412-0469/96Aerospatiale modifications: Mod 01100-<->->- Mod 01200-<->->- Mod 01300.Aircraft Flight Manual Supplement, ref: POL 000 M0001/C41Or later revisions of the above listed documents approved by (...) | N/A | 26.02.2010 |
| 10013957 | AIRBUS S.A.S. | EASA.A.S.00514 | REV. 3 | | A310-304 | - EASA STC EASA.A.S.00514 revision 3- The STC EASA.A.S.00514 allowed the installation of the Air to Air Refuelling kit provision, on the A310-304 aircraft model converted in a MultiRole Transport version, called #Polaris# configuration.- Revision 1 in (...) | - The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- See Certification Summary Document referenced in Associated Technical Documentation (...) | - Certification Summary Document, ref: A/BWT/C 412-0069/96 Engineering Order, ref: EO P2L 006/04A Issue 2, dated 17 January 2007- Aircraft Flight Manual Supplement Revision 9, Dated 13 February 2007- Aircraft Flight Manual Temporary Revision, ref: T (...) | N/A | 19.02.2010 |
| 10013960 | AIRBUS S.A.S. | EASA.A.S.00634 | REV. 6 | | A310-203/-221/-222 A310-304/-308/-324/-325 | EASA STC 10013960 Revision 6 (A310-200 series and -300 series aeroplanes, Passenger to Freighter conversion)(Ref. to former Luftfahrt Bundesamt STC TA0383 Issue 9)- The STC EASA.A.S.00634 has covered the transfer of the STC TA0383 Issue 9 (original Av (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. See Certification Summary Documents referenced in Associated Technical Documentation se (...) | For A310-200 series Aeroplanes- Certification Summary Document (except MSN 349): G22121315/000-001/COS Rev G, dated 10/02/2005 - Certification Summary Document (for MSN 349): EQM/001/000-003/COS, dated 16 April 2004 - Airbus Deutschland Engineering (...) | None | 27.06.2012 |
| 10013966 | AIRBUS S.A.S. | EASA.A.S.00685 | REV. 1 | | AIRBUS A321-231 | VIP cabin modification in accordance with Modification Approval Sheet (MAS) ref. SOG-792 issue 1, dated March 31st, 2005 | | Airplane Flight Manual Supplement (AFM-S) ref. DIT/B/C/1202/397/03 issue 1, dated January 18, 2005. Master Minimum Equipment List Supplement (M MEL-S) ref. '01-26 P5', '01-26 P5A', '02-26 P1', '01-33 P5', '01-33 P6', '01-35 P3', '01-52 P7', '02-52 P1'. | This STC is limited to A321-231 MSN 1202 | 04.06.2008 |
| 10013972 | AIRBUS S.A.S. | EASA.A.S.00767 | REV. 1 | | AIRBUS A320-214 | LIVE TV Installation | | Modification Approval Sheet (MAS) Ref.: SOG-00-91, Issue 1 dated May 27, 2005. MPD Section 9-2 Supplement Ref.: MPD-S MOD-SOG-00-901, dated May 27, 2005 | This STC is limited to A320-213 MSN 2165 | 04.07.2008 |
| 10013973 | AIRBUS S.A.S. | EASA.A.S.00769 | REV. 1 | | AIRBUS A320-214 | Avionic Rack 9100 VU Installation | | Modification Approval Sheet (MAS) Ref. SOG-00-902 Issue 1 dated May 27, 2005. AFM Supplement Ref. FM-S MOD-SOG-00-902 dated May 10, 2005. M MEL Supplement Ref.: M MEL-S MOD-SOG-00-902 dated May 1, 2005 | This STC is limited to A320-214 MSN 2165 | 04.07.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------|----------------|----------------|------------------------|--|--|---|--|--|---------------|
| 10013974 | AIRBUS S.A.S. | EASA.A.S.00772 | REV. 1 | | AIRBUS A320-214 | NAS VIP Cabin Modification | | Modification Approval Sheet (MAS) Ref.: SOG-00-900 Issue 1 dated June 1, 2005. AFM Supplement Ref.: FM-S MOD-SOG-00-900 Revision 3 dated May 27, 2005. MPD Section 9-2 Supplement, Ref.: MDP-S MOD-SOG-00-900 dated May 27, 2005. MMEL Supplement Ref.: MMEL-S MOD-SOG-00-900 Revision 00 dated May 1, 2005. | This STC is limited to A320-214 MSN 2165 | 04.07.2008 |
| 10014023 | AIRBUS S.A.S. | EASA.A.S.01078 | REV. 1 | | AIRBUS A319-133 | AL SALAM Master Modification | | MAS SOG-00-1030, Issue 1 dated October 4, 2005. AFM Supplement Ref FM-S MOD-SOG-00-1030 dated September 21, 2005. MPD Section 9-2 Supplement, Ref MDP-S MOD-SOG-00-1030, dated September 21, 2005. MMEL Supplement Ref. MMEL-S MOD-SOG-00-1030, dated September 21, 2005 | This STC is limited to A319-133 CJ MSN 2421 | 02.07.2008 |
| 10014127 | AIRBUS S.A.S. | EASA.A.S.01237 | REV. 1 | | AIRBUS A321-231 | Satcom in accordance with Morification Approval Sheet (MAS)ref. : SOG-00-794 issue 3, dated March 18, 2005. | | | This STC is limited to A321-231 MSN 1202 | 04.07.2008 |
| 10014140 | AIRBUS S.A.S. | EASA.A.S.01253 | REV. 1 | | AIRBUS A340-541 | DA WLAT Qatar Airways Modifications in accordance with Modification Approval Sheet Ref. SOG-01-0648 | | *Modification Approval Sheet, Doc Ref. SOG-01-0648, Issue 1 dated 17 March 2006 | The applicability of this STC is limited to MSN 495.No more than 10 Laptops equipped with WiFi system are allowed in thecabin | 04.07.2008 |
| 10014164 | AIRBUS S.A.S. | EASA.A.S.01296 | REV. 1 | | AIRBUS A340-541 | DA WLAT Qatar Airways Live TV Modifications in accordance with Modification Approval Sheet Ref. SOG-23-690 | | *Modification Approval Sheet, Doc Ref. SOG-23-690, Issue 1 dated 17 March 2006*Master Drawing List, Doc. Ref. MDL MOD SOG-23-690, Issue A*Airplane Flight Manual Supplement, Doc. Ref. FM-S MOD-SOG-23-690, Rev. 0 dated 21 September 2005* | The applicability of this STC is limited to MSN 495 | 04.07.2008 |
| 10014197 | AIRBUS S.A.S. | EASA.A.S.01340 | REV. 1 | | AIRBUS A330-200 | Installation of a more powerful air heater (3kg W) inside the private cabin washroom | | The associated Modification Approval Sheet is referenced SOG-21-1135 issue 2 dated 15/06/05. All documents related to this STC are listed in the Master Data List Drawing n°MDL-21-1135, Issue N° A (or later EASA approved revision). Technical Repercussion Sheet reference 21-1135 issue B | 1.This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer.2. The limitat (...) | 04.07.2008 |
| 10014231 | AIRBUS S.A.S. | EASA.A.S.01394 | REV. 1 | | AIRBUS A340-212 | VIP Cabin and systems upgrade in accordance with Sogema Modification Approval Sheet (MAS) SOG-00-0977 issue 3 dated 03/02/06 | | Flight Manual Supplement ref. FM-S MOD-SOG-F4-00-977 issue 6 dated 01/02/2006MPD Supplement ref. MPD-S SOG-F4-00-977 dated 31/01/2006(or later approved revision) | The approval of this modification is limited to the Airbus A340-212 MSN 061 | 04.07.2008 |
| 10014233 | AIRBUS S.A.S. | EASA.A.S.01396 | REV. 1 | | AIRBUS A319-133 | Avionics Rack 9100 VU installation | | - Modification Approval Sheet (MAS) ref. SOG-00-1056, issue1, dated October 4, 2005. AFM Supplement, ref. FM-S MOD-SOG-00-1056, dated September 21, 2005. MMEL Supplement, ref. MMEL-S MOD-SOG-00-1056, dated September 21, 2005 | 1. This STC is limited to A319-133 CJ MSN 2421 | 04.07.2008 |
| 10014234 | AIRBUS S.A.S. | EASA.A.S.01397 | REV. 1 | | AIRBUS A319-133 | Live TV system provision | | Modification Approval Sheet (MAS) ref SOG-00-1139, Issue 1, dated October 4, 2005 MPD Section 9-2 Supplement, ref. MPD-S MOD-MMEL-SMOD-SOG-001139, dated September 21, 2005 | This STC is limited to A319-133 CJ MSN 2421 | 04.07.2008 |
| 10014408 | AIRBUS S.A.S. | EASA.A.S.01703 | | | AIRBUS A300-600ST MODEL F4-608 | Installation of ISIS on A300-600STInstallation of two Integrated Stand-by Instrument System (ISIS) in replacement of the stand by speed, attitude and altitude. | | Change Approval Sheet, ref : ST34MOD015 Issue 1AFM Supplement ref : ST34MOD015, dated 19 October 2006 | This STC is approved only for the product configuration as defined in the approved design data referred to in paragraph#Description#. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 26.10.2006 |
| 10014419 | AIRBUS S.A.S. | EASA.A.S.01740 | REV. 1 | | AIRBUS A320 | SAT TV Tailwind 550 antenna installation in accordance withMAS SOG-53-1252 issue 2 dated 27/07/06 | | MPD Supplement MOD SOG-53-1252 dated Feb 01/06 | 1. The installation is compatible with the installation of a VH3 antenna between Fr 51 and Fr 52, and a Canadian Marconi Aero-H+ antenna between Fr 58 and Fr 61 (separately or simultaneously) 2. The installation requires both ACT 1 and 2 be install (...) | 30.06.2008 |
| 10014686 | AIRBUS S.A.S. | EASA.A.S.02849 | REV. 1 | | AIRBUS A319 | Modification MOD SOG-53-1296 * A319-SAT TV Tailwind 550 Antenna Installation* | | Associated Technical Documentation was approved under the STC EASA.A.S.01397 | | 30.06.2008 |
| 10014779 | AIRBUS S.A.S. | EASA.A.S.02977 | REV. 1 | LBA STC TA0015 ISSUE 4 | A300B2-1A/ -1C/ K-3C/ -203 A300B4-2C/ -103/ -203 | - EASA STC EASA.A.S.02977 revision 1- (Ref. to former Luftfahrt Bundesamt STC TA0015 Issue 4)- The STC EASA.A.S.02977 has covered the transfer of the STC TA0015 Issue 4 (original Aviation Authority reference) from the originalAviation Authority (LBA) (...) | - The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/approval- See Certification Summary Document referenced in Associated TechnicalDocumentation s (...) | - Certification Summary Document EQM 236/000-001/COS Issue 06.- Airbus Deutschland Engineering Order Nr. PFM 04296A Issue 04.- Aircraft Flight Manual Supplement for Airbus A300B2 / B4(->->)Revision 07, dated 15 July 2006.- Supplemental Airworthines (...) | N/A | 26.02.2010 |
| 10014911 | AIRBUS S.A.S. | EASA.A.S.03155 | | | AIRBUS A320-214 | Airbus Corporate Jet Centre Modification CJ0103 #VIP Interior - Shining Star No. 2 | | Modification Approval Sheet ref.: CJ0103 Issue 1, dated 25th February 2008AFM Supplement: FM-S MOD CJ0103, dated 25/02/08Instructions for Continued Airworthiness (ICA):MPD-S MOD CJ0103, dated 25th February 2008AMM-S MOD CJ0103, dated 01st November 2007 | | 03.03.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------|------------------|----------------|------------------------------|--------------------|---|---|--|--|---------------|
| 10015025 | AIRBUS S.A.S. | EASA.A.S.03332 | | | AIRBUS A320-214 | Airbus Corporate Jet Centre Modification CJ0145 Issue 1 - VIP Interior - Shining Star No. 1 | | Modification Approval Sheet ref.: CJ0145 Issue 1, dated 12 December 2008AFM Supplement Ref: FM-S MOD CJ0145, Revision 00, dated 24/11/08MPD Supplement Ref: MPD-S MOD CJ0145 dated 01/12/08MPD Supplement SAT TV Ref: MPD-S MOD CJ0277 dated 01/10/08 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 12.12.2008 |
| 10015053 | AIRBUS S.A.S. | EASA.A.S.03374 | | | AIRBUS A319-115 | Airbus Corporate Jet Centre Modification CJ0222 Issue 1VIP Interior - Residence | | Modification Approval Sheet ref.: CJ0222 Issue 1, #Top Modification Residence#, dated 19th June 2009AFM Supplement Ref: FM-S MOD CJ0222, Revision 00, dated 1st March 2009MPD Supplement Ref: MPD-S MOD CJ0222 dated 1st March 2009MPD Supplement Ref: MPD-S MOD CJ0310 dated 1st March 2009 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 22.06.2009 |
| 10015198 | AIRBUS S.A.S. | EASA.A.S.03589 | | | AIRBUS A320-214 | Airbus Corporate Jet Centre Modification CJ0356 Issue 1 #AFM Supplement - VIP Interior: Shining Star No. 2 | | Modification Approval Sheet ref.: CJ0356 Issue 1, dated 06th June 2008AFM Supplement: FM-S MOD CJ0103, dated 06/06/08 | | 06.06.2008 |
| 10015210 | AIRBUS S.A.S. | EASA.A.S.03625 | | DGAC STC 180-SF-0227 | AIRBUS A320 SERIES | Installation of a cabin VIP (Aeroservices) in accordance with the Service Bulletin SOG-WD-00-00-01 (or later approved revision) | | Supplement no 180-SF-0227 to data sheet no 180Service Bulletin SOG-WD-00-00-01 (or later approved revision) | | 04.07.2008 |
| 10015211 | AIRBUS S.A.S. | EASA.A.S.03626 | | DGAC FRANCE 180-SF-0187 | AIRBUS A320 SERIES | Cabin Conversion of one A319 corporate jet (A319 CJ) | | - Sogerma document no SOG-01-128 issue 3, dated February 22, 2002 | | 07.07.2008 |
| 10015212 | AIRBUS S.A.S. | EASA.A.S.03627 | | DGAC FRANCE STC 180-SF-0212 | AIRBUS A320 SERIES | Cabin Conversion of one A319 Corporate Jet (A319 CJ) | | Type Certificate Data Sheet no 180 and its supplement no 180-SF-0212, with deviations identified and listed in the Certification Basis document, CRI G-4102 issue 3, dated November 26, 2003Sogerma document no SOG-01-275 issue 5, dated November 27, 2003 | | 07.07.2008 |
| 10015281 | AIRBUS S.A.S. | EASA.A.S.03813 | | LBA STC NO.2830/0009 ISSUE 1 | A300C4-203 | - Ref. to former Luftfahrt Bundesamt STC 2830/0009 Issue 1 - This EASA STC keeps the certificated design of the LBA STC 2830/0009 Issue 1, which consisted in the installation of structure and system provisions to change the loading system to an integrat (...) | - The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- See Certification Summary Document referenced in Associated Technical Documentation (...) | - Daimler-Benz Aerospace Airbus Engineering Order Nr. SAA 03696A Issue 1, dated 30 August 1996. Or later revisions of the above listed documents approved by EASA | Limited to MSN 212 | 26.02.2010 |
| 10015282 | AIRBUS S.A.S. | EASA.A.S.03814 | | LBA STC NO.TA0466 | A300F4-605R | - Ref. to former Luftfahrt Bundesamt STC TA0466- This EASA STC keeps the certificated design of the LBA STC TA0466, which consisted in the retrofit of the Lower Deck Cargo Compartment (LDCC) - Cargo Loading System (CLS), and reflects the changes induced (...) | - The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- The certificated noise and/ or emissions levels of the original product are unchanged a (...) | - Certification Summary Document : EASA/243301/255/CS3 Issue 01 - Airbus Engineering Order Nr. FDX 128/99A, Issue 06 Or later revisions of the above listed documents approved by EASA | Limited to the aircraft model operated by Federal Express (FedEx) | 26.02.2010 |
| 10015283 | AIRBUS S.A.S. | EASA.A.S.03816 | | LBA STC NO. TA0217 | AIRBUS A319-132 | Equipment / Furnishing - Passenger Compartment - VIP-Kit for A319(Khara) Group Kuwait Revision to Luftfahrt Bundesamt STC TA0217 Issue 2. This EASA STC keeps the certificated design of the LBA STC TA0217 issue 2, which consisted in the aircraft VIP cabi (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. See Engineering Order referenced in Associated Technical Documentation section. The cert (...) | DASA Engineering Order Nr. K9G003/99A, Issue 2 or later revisions of the above listed documents approved by EASA. | Limited to MSN 913. | 12.05.2010 |
| 10015284 | AIRBUS S.A.S. | EASA.A.S.03817 | | LBA STC NO. TA0265 | AIRBUS A319-133 | Equipment / Furnishings - Passenger Compartment - Install cabin furnishing for Daimler Chrysler Aviation Revision to Luftfahrt Bundesamt STC TA0265 This EASA STC keeps the certificated design of the LBA STC TA0231 which consisted in the aircraft cabin (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. See Engineering Order referenced in Associated Technical Documentation section. The cert (...) | DASA Engineering Order Nr. L4C015/99A, Issue 1 or later revisions of the above listed documents approved by EASA | None. | 12.05.2010 |
| 10015285 | AIRBUS S.A.S. | EASA.A.S.03818 | | LBA STC NO. TA0269 | AIRBUS A319-133 | Equipment / Furnishing - Passenger Compartment - Install baggage container for DaimlerChrysler Aviation. Revision to Luftfahrt Bundesamt STC TA0269 Issue 2. This EASA STC keeps the certificated design of the LBA STC TA0269 Issue 2 which consisted in the (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. See Engineering Order referenced in Associated Technical Documentation section. The cert (...) | DASA Engineering Order Nr. C8A00100A, Issue 2 or later revisions of the above listed documents approved by EASA | Limited to MSN 1053 | 12.05.2010 |
| 10015286 | AIRBUS S.A.S. | EASA.A.S.03819 | | LBA STC NO. TA0231 | AIRBUS A320-232 | Equipment/ Furnishings - Passenger Compartment - VIP-kit for A320 Syria. Revision to Luftfahrt Bundesamt STC TA0231. This EASA STC keeps the certificated design of the LBA STC TA0231 which consisted in the aircraft cabin VIP reconfiguration, and reflect (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Deviation to the Certification requirements :- JAR 25.785 h)1), - JAR 25.853a) - JAR 25. (...) | DASA Engineering Order Nr. SYR037/98A or later revisions of the above listed documents approved by EASA | Limited to MSN 886, 918, 1032, and 1075 | 12.05.2010 |
| 10015287 | AIRBUS S.A.S. | P-EASA.A.S.03820 | | LBA STC NO. TA0228 | A330-223 | (Ref. to former Luftfahrt Bundesamt STC TA0228) This EASA STC keeps the certificated design of the LBA STC TA0228 which consisted in the aircraft cabin reconfiguration, and reflects the changes induced by the transfer of LBA STC TA0228 from DaimlerChrysler Aerospace to Airbus SAS. | The Certification Basis for the original product remains applicable to this certificate/ approval. See Certification Program Document referenced in Associated Technical Documentation section. The certificated noise and/ or emissions levels of the original (...) | DASA Engineering Order Nr. SWR010/99A Issue 3 or later revisions of the above listed documents approved by EASA | Limited to MSN 240, 249, 253, 255, 262, 275, 288, 291, and 366 | 31.08.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------|------------------|----------------|----------------------|--------------------|--|---|--|---|---------------|
| 10015288 | AIRBUS S.A.S. | EASA.A.S.03821 | | LBA STC TA0261 | AIRBUS A310-304 | GLARE demonstrator panel installation Revision to Luftfahrt Bundesamt STC TA0261. This EASA STC keeps the certified design of the LBA STC TA0261, which consisted in the installation of the Glare demonstrator panel for testing purpose, and reflects th (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. See Certification Program Document referenced in Associated Technical Documentation. The certified noise and/ or emissions levels of the original (...) | Certification Program : EAA484/530/001/GLARE, dated 17 March 1999- Glare panel installation - Daimler-Chrysler Aerospace/Airbus Engineering Order Nr. GAF0689A, Issue 02- Glare panel replacement - Airbus Engineering Order Nr. GAF030/08A, Issue 01 or late (...) | Limited to MSN 484. | 12.05.2010 |
| 10028319 | AIRBUS S.A.S. | P-EASA.A.S.03500 | REV. 1 | | A319-115, MSN 3243 | ACJ Centre Modification CJ0300 Issue 1 (VIP Cabin Interior - ALF) - ACJ Centre Modification CJ0392 Issue 1 (Private to Public Conversion) | In accordance with the EASA TCDS A.064 plus. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval - Certification Review Item CRI G-4110 "ACJ (...) | 1. Modification Approval Sheet ref: CJ0300 Issue 1 "Top Modification ALF" dated 16th December 20092. Modification Approval Sheet ref: CJ0392 Issue 1 "Conversion Private to Public" dated 16th December 20093. AFM Supplement Ref: FM-S MOD CJ0300, dated 1 (...) | This STC is limited to Airbus A319 MSN 3243 | 17.12.2009 |
| 10028791 | AIRBUS S.A.S. | EASA.A.S.00351 | REV. 2 | | A310-304 | - EASA STC EASA.A.S.00351 revision 2 (Ref. to former Luftfahrt Bundesamt STC TA0175 Issue 3) - The STC EASA.A.S.00351 has covered the transfer of the STC TA 0175 Issue 3 (Original Aviation Authority reference) from the original Aviation Authority (LBA) 1 (...) | - The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval - See Certification Summary Document referenced in Associated Technical Documentation (...) | - Certification Summary Document EQM 434/000/001/CO2 Issue 03 - Airbus Deutschland Engineering Order Nr. GAF 035/96A Issue 04, dated 14 December 2006 - Aircraft Flight Manual Supplement for Airbus A310-304, Revision 09, dated 24 January 2007 Or later (...) | 1- Transmissions (transmission operation) in tactical/close air support band (frequency range 30-89 MHz) are prohibited on the military radios/V/UHF Cockpit and V/UHF FOS during flight. 2- Aircraft Flight Manual Supplement for Airbus A310-304, Temporary (...) | 19.02.2010 |
| 10028999 | AIRBUS S.A.S. | P-EASA.A.S.03735 | REV. 1 | | A319-115 | ACJ Centre Modification CJ0400 Issue 1 (VIP Cabin Interior - ACROPOLIS) | In accordance with the EASA TCDS A.064 plus. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval - Certification Review Item CRI G-4111 "ACJ (...) | 1. Modification Approval Sheet ref: CJ0400, Issue 1 "Top Modification ACROPOLIS" dated 24th February 20102. AFM Supplement Ref: FM-S MOD CJ0400, Issue 00, dated 1st February 20103. MPD Supplement Ref: MPD-S MOD CJ0400, Issue 00, dated 1st February 2010 (...) | This STC is limited to Airbus A319 MSN 3826 | 25.02.2010 |
| 10030147 | AIRBUS S.A.S. | P-EASA.A.S.03949 | | | A320-200 | Airbus Corporate Jet Centre Modification CJ0531 Issue 1 - VIP KIT - RAFO Project | In accordance with the EASA TCDS A.064 plus. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval - Certification Review Item CRI A-2301-003 (...) | 1. Modification Approval Sheet ref: CJ0531 Issue 1 <-> dated 27th May 20102. AFM Supplement Ref: FM-S MOD CJ0531, dated May 01/10 or later revisions of the above listed documents approved by EASA | This STC is limited to Airbus A320-200 MSN 3723 & MSN 4117 | 28.05.2010 |
| 10031829 | AIRBUS S.A.S. | P-EASA.A.S.03426 | REV. 1 | | A310-304 | Replacement of Cockpit Communication Unit (CCU) and Radio Control Unit (RCU) by Centralized Radio Control (CRC). This STC will modify the communication system by changing equipments, mating connectors, and wiring. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval - Special Condition: H-01 Issue 1 (Enhanced Airworthiness Programme for Aeroplane System (...) | Master Document List, Ref: EAWG-027-1/10/08, Issue A Change Proposal, ref: P2L_06704A, Issue 1 Change Proposal, ref: P2L_06804A, Issue 1 Aircraft Flight Manual Supplement, dated 15 September 2010. Instruction for Operation: A310-304 Canadian Armed Force (...) | The current STC needs specific aircraft configuration as a prerequisite. To install this STC on A310-304 aircraft models, embodiment of EASA STC ref: 10013946, Rev 1 - previously referenced as EASA STC EASA.A.S.00403 - (or any further EASA approved revision) is required. | 09.12.2010 |
| 10031840 | AIRBUS S.A.S. | P-EASA.A.S.03950 | | | A320-232 | Airbus Corporate Jet Centre Modification CJ0540 Issue 1 - VIP Interior - "QATAR" | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval - CRI A-2301-002 "ACJC Centre Conversion Project Certific (...) | Modification Approval Sheet ref.: CJ0540 Issue 1, dated 17th September 2010 AFM Supplement Ref : FM-S MOD CJ0540, Revision 00, dated 1st August 2010 MPD Supplement Ref : MPD-S MOD CJ0540, dated 1st August 2010 MPD Supplement SAT TV Ref : MPD-SMOD CJ0277 (...) | This STC is limited to A320-232 MSN 4170 | 17.09.2010 |
| 10033449 | AIRBUS S.A.S. | | | | A319-133 | Airbus Corporate Jet Centre Modification CJ0615 Issue 1 - VIP Interior - "HNA" | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval - CRI G-4113 "ACJC Centre Conversion Project Certific (...) | - Modification Approval Sheet ref.: CJ0615 Issue 1, dated 20th January 2011 - AFM Supplement Ref : FM-S MOD CJ0615, Revision 00, dated 1st January 2011 - MPD Supplement Ref : MPD-S MOD CJ0615, dated 1st January 2011 or later revisions of the above listed documents approved by EASA | This STC is limited to A319-133 MSN 4042 | 21.01.2011 |
| 10034553 | AIRBUS S.A.S. | P-EASA.A.S.03845 | | | A319-115 | Airbus Corporate Jet Centre Modification CJ0500 Issue 1 - VIP Interior - "ORION" | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval - CRI G-4112 "ACJC Centre Conversion Project Certific (...) | - Modification Approval Sheet ref: MAS CJ0500 Issue 1 "TOP MODIFICATION ORION" dated 11/04/2011;- Modification Approval Sheet ref: MAS CJ0526 Issue 1 "CONVERSION PRIVATE / PUBLIC" dated 11/04/2011;- AFM Supplement ref: FM-S MOD CJ0500 dated 1st of Mar (...) | This STC is limited to A319-115 MSN 3994 | 12.04.2011 |
| 10036239 | AIRBUS S.A.S. | | | | A320-200 | Airbus Corporate Jet Centre Modification CJ1000 Issue 1 - VIP Interior - "RAFO 3" | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval - CRI A 2301-004 "ACJC Centre Conversion Project Certific (...) | - Modification Approval Sheet, ref.: CJ1000, Issue 1, dated 23rd August 2011 - AFM Supplement, ref.: FM-S MOD CJ1000, Revision 00, dated August 2011 - Conversion Instruction, ref.: CI 00CJ007, Revision A, dated 25th July 2011 or later revisions of the above listed documents approved by EASA. | This STC is limited to A320-214 MSN 4795. | 24.08.2011 |
| 10037015 | AIRBUS S.A.S. | | | | A319-133 | Airbus Corporate Jet Centre Modification CJ0700 Issue 1 - VIP Interior - "OPTIMA PROJECT" | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval - CRI A 4301-004 "ACJC Centre Conversion Project Certific (...) | - Modification Approval Sheet, ref. CJ0700, Issue 1, dated 25th October 2011 - Airplane Flight Manual - AFM - Supplement, ref.: FM-S MOD CJ0700, Revision 00, dated August 2011 - Maintenance Planning Document - MPD - Supplement, ref. MPD-S MODCJ0700, Re (...) | This STC is limited to A319-133 MSN 4319 | 25.10.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|----------------|----------------|----------------------|--------------------------------|---|---|---|---|---------------|
| 10037612 | AIRBUS S.A.S. | | | | A320-214 | Airbus Corporate Jet Centre Modification CJ1118 Issue 1 - Cabin System Upgrade | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- CRI A-2301-005 "ACJC Centre Conversion Project Certific (...) | - Modification Approval Sheet ref.: CJ1118 Issue 1, dated 9th December 2011- Airplane Flight Manual Supplement, ref.: FM-S MOD CJ1118, dated 1st December 2011 or later revisions of the above listed documents approved by EASA | This STC is limited to A320-214 MSN 2165 | 12.12.2011 |
| 10037703 | AIRBUS S.A.S. | | | | A319-115 | Airbus Corporate Jet Centre Modification CJ0800 Issue 1 - Project "2A" | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- CRI A-4301-005 "ACJC Centre Conversion Project Certific (...) | - Modification Approval Sheet ref.: CJ0800 Issue 1, dated 16th December 2011- Modification Approval Sheet "Conversion Private/Public and Vice Versa" ref.: CJ0875 Issue 1, dated 16th December 2011- Airplane Flight Manual Supplement, ref.: FM-S MOD CJ0 (...) | This STC is limited to A319-115 MSN 4470 | 16.12.2011 |
| 10038826 | AIRBUS S.A.S. | | | | A319-115 | Airbus Corporate Jet Centre Modification CJ0950 Issue 1 - VIP Interior - "Metropole". | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- CRI A 4301-010 "ACJC Centre Conversion Project Certification Basis". The requirements (...) | - Modification Approval Sheet, ref.: CJ0950, Issue 1, dated 19th March 2012;- Airplane Flight Manual AFM Supplement, ref.: FM-S MOD CJ0950, Revision 00, dated 1st February 2012;- Maintenance Planning Document MPD Supplement, ref.: MPD-S MOD CJ0950, date (...) | This STC is limited to A319-115 MSN 4024 and MSN 4679. | 19.03.2012 |
| 10039051 | AIRBUS S.A.S. | | | | A321-200 | Airbus Corporate Jet Centre Modification CJ1094 Issue 1 - Installation of Business Class Seating Area - "Gulf Air (Type C)" | In accordance with the EASA TCDS. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- CRI A 3301-001 "ACJC Centre Conversion Project Cer (...) | - Modification Approval Sheet, ref.: CJ1094, Issue 1, dated 2nd April 2012;- Airplane Flight Manual AFM Supplement, ref.: FM-S MOD CJ1094, dated 1st April 2012; or later revisions of the above listed documents approved by EASA. | None. | 03.04.2012 |
| 10039666 | AIRBUS S.A.S. | | | | A320-214 | Airbus Corporate Jet Centre Modification CJ1100 Issue 1 - Installation of Business Class Seating Area - "Gulf Air - Type A" | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CRI A 2301-006 "ACJC Centre Conversion Project Certific (...) | - Modification Approval Sheet, ref.: CJ1100(->->-) Issue 1, dated 16th May 2012- Airplane Flight Manual AFM Supplement(->->-) ref.: FM-S MOD CJ1100, dated 1st May 2012 or later revisions of the above listed documents approved by EASA | This STC is limited to MSN 4827, MSN 4860, MSN 4865 | 16.05.2012 |
| 10040777 | AIRBUS S.A.S. | | | | A320-214 | Airbus Corporate Jet Centre Modification CJ1108 Issue 1 - Installation of Business Class Seating Area - "Gulf Air - Type B" | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CRI A 2301-007 "ACJC Centre Conversion Project Certific (...) | - Modification Approval Sheet, ref.: CJ1108, Issue 1, dated 25th July 2012- Airplane Flight Manual AFM Supplement, ref.: FM-S MOD CJ1108, dated 1st July 2012 or later revisions of the above listed documents approved by EASA | This STC is limited to MSN 5171, MSN 5175 | 26.07.2012 |
| 10044692 | AIRBUS S.A.S. | | | | A319-115 | Airbus Corporate Jet Centre Modification CJ0900 -VIP Project "UAE - EMIRATES" | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements as listed in the General CRI A-4301-011 "ACJC Project Certification Basis"- Special Condition(s): CRI E-21 - Isolated Comp (...) | - Modification Approval Sheet, ref.: CJ0900, Issue 1, dated 29th April 2013- Airplane Flight Manual - AFM Supplement, ref.: FM-S MOD CJ0900, dated 1st March 2013- Maintenance Planning Document - MPD Supplement, ref.: MPD-S MODCJ0900, dated 1st March 2 (...) | This STC is limited to MSN 4822. Structural Inspections Tasks of the antennas have been requested as per the installation of the ALS of the referenced MPD Supplement, ref. MPD-S MODCJ0900. Prior to installation of this design change it must be determined that the int (...) | 29.04.2013 |
| 10045983 | AIRBUS S.A.S. | | | | A319-115 | Airbus Corporate Jet Centre Modification CJ1620 - "MASTER MOD CJ -RELIANCE" | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements as listed in the General CRI A-4301-019 "ACJC Project Certification Basis". Special Condition(s): CRI H-01 - ICAo n EWIS The (...) | Modification Approval Sheet, ref.: CJ1620, Issue 1, dated 30. July 2013 Airplane Flight Manual - AFM Supplement, ref. FM-S MOD CJ1620, dated 1. July 2013 Maintenance Planning Document - MPD Supplement, ref. MPD-S MOD CJ1620, dated 28. June 2013 or later (...) | This STC is limited to MSN 2837. Prior to installation of this design change it must be determined that the interrelationship between this design change and/ or previously installed design change and/ or repair will introduce no adverse effect upon (...) | 01.08.2013 |
| 10046179 | AIRBUS S.A.S. | | | | A320-200 | Airbus Corporate Jet Centre Modification CJ1467 - "MASTER MOD CJ - RAFO (Cabin & Systems Upgrade)" | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements as listed in the General CRI A-2301-009 "ACJC Project Certification Basis". Special Condition(s): CRI H-01 - ICAo n EWIS Devi (...) | - Modification Approval Sheet, ref.: CJ1467, Issue 1, dated 26th August 2013- Airplane Flight Manual - AFM Supplement, ref.: FM-S MOD CJ1467, dated 1st August 2013 or later revisions of the above listed documents approved by EASA. | This STC is limited to MSN 3723, MSN 4117 & MSN 4795. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce n (...) | 26.08.2013 |
| 10046468 | AIRBUS S.A.S. | | | | A319-115 | Airbus Corporate Jet Centre - Top Modification CJ1200 - VIP Project "BELAIR" | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements as listed in the General CRI A-4301-012 "ACJC Project Certification Basis" in Appendix to this TV. The requirements for envi (...) | Modification Approval Sheet, ref. CJ1200, Issue 1, dated 20th September 2013 Airplane Flight Manual - AFM Supplement, ref.: FM-S MOD CJ1200 Revision 1st June 2013 Maintenance Planning Document - MPD Supplement, ref.: MPD-S MODCJ1200 Revision 1st June 2 (...) | This STC is limited to MSN 4228. | 23.09.2013 |
| 10014094 | AIRBUS TRANSPORT INTERNATIONAL SNC | EASA.A.S.01192 | | | AIRBUS A300-600ST MODEL F4-608 | STC ST46MOD001 Issue 01: A300-600ST / Data Link System. This controller data link communication based on VDL mode 2 project is led by Eurocontrol. The crew will be able to exchange data and messages with a ground-based station airline operation centre and air traffic services. | | Change Approval Sheet No ST46MOD001 issue 01. Supplement Aircraft Flight Manual A300-600ST, dated 24/03/2005. | This STC is approved only for A300F4-608ST. | 12.04.2005 |
| 10014202 | AIRBUS TRANSPORT INTERNATIONAL SNC | EASA.A.S.01352 | | | AIRBUS A300-600ST MODEL F4-608 | ST34MOD027 "Enhanced Surveillance on A300-600ST" | | Supplement to A300-600ST Flight Manual referred ST34MOD027 dated 28 March 2006, or any later EASA approved revision | | 07.04.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|------------------|----------------|----------------------|--------------------------------|--|---|---|--|---------------|
| 10014921 | AIRBUS TRANSPORT INTERNATIONAL SNC | EASA.A.S.03168 | | | AIRBUS A300-600ST MODEL F4-608 | Implementation of Automatic Dependent Surveillance # Broadcast in Non Radar Airspace capability (ADS-B NRA) on A300-600ST. This STC provides surveillance services in non-radar airspaces using ADS-B Out. ADS-B NRA is part of the Eurocontrol #Cascadef project. | | - Change Approval Sheet, ref: ST34MOD030 Issue 1 (Any later approved revision). - AFM Supplement ref: ST34MOD030, dated 12 November 2008. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 22.12.2008 |
| 10015056 | AIRBUS TRANSPORT INTERNATIONAL SNC | EASA.A.S.03378 | | | AIRBUS A300-600ST MODEL F4-608 | FMS SwitchingThe modification installs the coupling of eachFMC to both FMS CDU and EFIS displays, in order to comply with the B-RNAV requirements in case of one FMC source failure. This modification allows dispatching the serviceable FMC signal on 1 (...) | | Fiche de Modification n° ST34MOD021, Issue 1 (or any later approved revision)MMEL Supplement: Annexe 7 of the Fiche de Modification n° ST34MOD021. | 1. The MMEL Supplement associated with this STC is only quoted for information. The STC approval EASA.A.S.03378 does not constitute approval. The MMEL Supplement approval will be granted separately.2. Prior to installation of this modification th (...) | 12.03.2008 |
| 10027571 | AIRBUS TRANSPORT INTERNATIONAL SNC | P-EASA.A.S.03848 | | | AIRBUS A300F4-608ST | Special Transports / Transports Spéciaux (This STC, ref. ST25MOD020ed.01 (Applicant reference), has been developed, reviewed, tested and found compliant with the Certification Basis to deal with non-Airbus sub-assemblies transport. On case by case basis (...) | As per EASA TCDS A.014 + CRI D-10 Issue 01.The Certification Basis for the original product and the followingadditionalor alternative airworthiness requirements are applicable tothis certificate/ approval. The certificated noise and/ or emissionslev (...) | Change Approval Sheet, ref ST25MOD020, Issue 48 or later revisions ofthis document approved by EASA. | | 12.10.2009 |
| 10028350 | AIRBUS TRANSPORT INTERNATIONAL SNC | | | | A300F4-608ST | Special Transports / Transports SpéciauxThis STC, ref. ST25MOD020 ed.01 (Applicant reference), has been developed, reviewed, tested and found compliant with the CertificationBasis to deal with non-Airbus sub-assemblies transport. On case bycasebasis, (...) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvals per EASA TCDS A.014 + CRI F-06 Issue 01The certificated noise and/ or emissions leve (...) | Change Approval Sheet, ref: ST25MOD020<->-> Issue 49 | None | 18.12.2009 |
| 10032203 | AIRBUS TRANSPORT INTERNATIONAL SNC | | | | A300F4-608ST | Special Transports / Transports SpéciauxThis STC, ref. ST25MOD020 ed.01 (Applicant reference), has been developed, reviewed, tested and found compliant with the CertificationBasis to deal with non-Airbus sub-assemblies transport. On case bycasebasis, (...) | The Certification Basis for the original product remains applicable tothis certificate/ approvalThe requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval | Change Approval Sheet, ref : ST25MOD020<->-> Issue 50or later revisions of the above listed documents approved by EASA | None | 15.10.2010 |
| 10037753 | AIRBUS TRANSPORT INTERNATIONAL SNC | | | | A300F4-608ST | Special Transports / Transports SpéciauxIssue 1 of ATI STC ref. ST25MOD020 (Applicant reference)<->->was developed, reviewed, tested and found compliant with the CertificationBasis to deal with non-Airbus sub-assemblies transport. On case by case (...) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval- Cf EASA TCDS A.014- CRI F-06 Issue 2 : Transport of Instrumented Payloads powered by (...) | Change Approval Sheet and Fiche de Modification ref : ST25MOD020, Issue51 or later revisions of the above listed documents approved by EASA | None | 20.12.2011 |
| 10040583 | AIRBUS TRANSPORT INTERNATIONAL SNC | | | | A300F4-608ST | Special Transport - Instrumented Airbus Sectionsissue 1 of ATI modification ST25MOD020 (applicant#s reference)<->->was developed, reviewed, tested and found compliant with theCertification Basis to deal with non-Airbus sub-assemblies transport. O (...) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval- Cf EASA TCDS A.014- CRI F-06 Issue 2 : Transport of Instrumented Payloads powered by (...) | Change Approval Sheet and Fiche de Modification ref : ST25MOD020, Issue52a or later revisions of the above listed documents approved by EASA | none | 11.07.2012 |
| 10040859 | AIRBUS TRANSPORT INTERNATIONAL SNC | | | | A300F4-608ST | Installation of an Indium Satcom for voice and ACARS on A300-600ST | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Change Approval Sheet ST23MOD021<->-> rev 1, date 30 July2012Airplane Flight Manual Supplement as in, ST23MOD021 Annex 7, rev 1, date30 July 2012or later revisions of the abovelisted documents approved by EASA inaccordance with EASA EDDecision 2004/04/CF (or subsequent revisions ofthis decision | None | 31.07.2012 |
| 10043319 | AIRBUS TRANSPORT INTERNATIONAL SNC | EASA.A.S.03751 | | | AIRBUS A300-600ST MODEL F4-608 | Installation of pressure seals on Frame (FR) 10.7. This STC consists in installing two pressure seals on FR10.7 in order toallow future avionicsinstallations. | | Change Approval Sheet, ref: ST53MOD011 Issue 1(Any later approved revision). | 1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. This Cer (...) | 31.07.2009 |
| 10045863 | AIRBUS TRANSPORT INTERNATIONAL SNC | | | | A300F4-608ST | Modification ST25MOD069 issue 1, "Installation of a 6th seat in the cockpit". This approval covers also approval of the following modifications, ST35MOD006 issue 1 "Oxygen supply for 5th and 6th occupant"; ST53MOD014"FR10.7, reinforcement of the pressur (...) | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): CRI D-301 OccupancyThe requirements for environmental protection and the associatedcertificated noise (...) | Airbus Transport International AFM supplement ref. AFM DO/2013/116 issueAcr later revisions of the above listed documents approved by EASA. | - The STC is limited to two Airbus A300F4-608ST model aeroplanes, Manufacturer Serial Number (MSN) 765 and 655. - Aeroplanes converted by this STC must be operated in accordance with the latest approved copy of the following documents: Airbus Transport (...) | 24.07.2013 |
| 10015032 | AIRCAD AIRCRAFT CABIN DESIGN | EASA.A.S.03344 | | | EMBRAER EMB-120 | EMB-120 VIP Cabin Modification is a major change designed by AIRCAD that converts half part of the airliner cabin to a VIP cabin with six VIP seats and two tables on pallets and a curtain dividing the airline and VIP areas. One of the three emergency exits is blocked. | | AIRCAD Certification Program CP 07-001, AIRCAD Service Bulletin ACD-SB 07-001. | None. | 11.07.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|--------------|----------------|----------------------|---|--|---|---|--|---------------|
| 10040380 | AIRCAD AIRCRAFT CABIN DESIGN | | | | F900EX | Partial Cabin Entertainment and Galley equipment upgrade | The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | File ref.: 12006 Certification Program CP-12006 Revision A, dated 27/06/2012. Service Bulletin ACD-SB-12006 Revision A, dated 27/06/2012, or later revisions of the above listed documents approved by EASA | Falcon 900EX S/N 054 | 28.06.2012 |
| 10040624 | AIRCAD AIRCRAFT CABIN DESIGN | | REV. 1 | | F2000 | Installation of a sliding door, main entryway door removable workshelves, installation or replacement of galleys equipment and IFE equipment, RH and annex RH galleys shelves rearrangement. | The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Certification Program CP-12001 Revision A, dated 13 July 2012; Service Bulletin ACD-SB-12001 Revision B; or later revisions of the above listed documents approved by EASA. | Falcon 2000 S/N 111. | 19.07.2012 |
| 10046112 | AIRCAD AIRCRAFT CABIN DESIGN | | | | A330-243 | A330-200 MSN 627 Cabin Reconfiguration - new seats and new IFE system installation. | The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Master Document List MDL-12007_Rev9, dated 24/07/2013; Instructions for Continued Airworthiness ICA-12007_Rev0, dated 07/02/2013; Aircraft Maintenance Manual Supplement AMM-12007_Rev0, dated 04/02/2012; or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 15.08.2013 |
| 10046236 | AIRCAD AIRCRAFT CABIN DESIGN | | REV. 1 | | MF50 | Thrane & Thrane TT-5006A IGA Satcom antenna installation Description as per Associated Technical Documentation | The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Master Data List Ref. MDL-13002 Revision 2, dated 31/07/2013. Certification Program CP-13002 Revision 1, dated 31/07/2013. ICA Manual ICA-13002 Revision 0, dated 10/07/2013. AMM Supplement AMMS-13002 Revision 0, dated 03/07/2013. Change Approval Sheet C (...) | None | 24.09.2013 |
| 10038356 | AIRCRAFT CABIN MAINTENANCE GmbH | | | | A318, A319, A320, A321 | Installation of refurbished Passenger Seat Belt P/N EPA20511-2-01-33 on 16g seats | CS-25 The Certification Basis for the original product remains applicable to this certificate/approval. | Refer to Master Document List E/O No. 62-146169 Rev. 0 (16.12.2011) or later revisions of the above listed documents approved by EASA | As per Service Bulletin No. 62-146169 SEB 06 00, latest revision. | 08.02.2012 |
| 10039575 | AIRCRAFT CABIN MAINTENANCE GmbH | | REV. 1 | | 1. A330 2. B737 3. B767 | Installation of refurbished Passenger Seat Belt P/N EPA20511-2-01-20 on 16g seats | CS-25 The Certification Basis for the original product remains applicable to this certificate/approval. | Refer to Master Document List E/O No. 82-1000845 Rev. 0 (11.04.2012) or later revisions of the above listed documents approved by EASA | As per Service Bulletin No. 82-1000845 SEB 06 00, latest revision | 21.05.2012 |
| 10042194 | AIRCRAFT CABIN MAINTENANCE GmbH | | | | SEE APPROVED MODEL LIST | Installation of refurbished Passenger Seat Belt P/N EPA 6488-2-01-40 on 16g seats | The Certification Basis for the original product remains applicable to this certificate/approval. | Refer to Master Document List E/O No. 104-1001541 Rev. 0 (12.11.2012) or later revisions of the above listed documents approved by EASA | As per Service Bulletin No. 104-1001541 SEB 06 00, latest revision | 14.11.2012 |
| 10044062 | AIRCRAFT CABIN MAINTENANCE GmbH | | | | SEE EASA APPROVED MODEL LIST | Installation of refurbished Passenger Seat Belt P/N EPA 27511-2-01-70 on 16 g seats | The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Refer to Master Document List E/O No. 112-1003111 Rev. 0 (27.02.2013) or later revisions of the above listed documents approved by EASA | As per Service Bulletin No. 112-1003111 SEB 06 00, latest revision Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair w (...) | 14.03.2013 |
| 10044880 | AIRCRAFT CABIN MAINTENANCE GmbH | | | | 1- A318-111, -112, -121, -122 1- A319-111, -112, -113, -114 1- A319-115, -131, -132, -133 1- A320-111, -211, -212, -214 1- A320-215, -216, -231 1- A320-232, -233 1- A321-111, -112, -131, -211 1- A321-212, -213, -231, -232 2- 737-100, -200, -200C, -300 2- 737-400, -500, -600, -700 2- 737-800, -900, -900ER | Installation of refurbished Passenger Seat Belt P/N EPA 20031-2-01-60 on 16g seats. | The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Refer to Master Document List E/O No. 116-1003350 Rev. 0 (29.04.2013) or later revisions of the above listed documents approved by EASA. | As per Service Bulletin No. 116-1003350 SEB 06 00, latest revision. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair (...) | 14.05.2013 |
| 10045658 | AIRCRAFT CABIN MAINTENANCE GmbH | | | | 737-100, 737-200, 737-200C, 737-300, 737-400, 737-500, 737-600, 737-700, 737-800, 737-900, 737-900ER. | Installation of refurbished Passenger Seat Belt P/Ns 27741-2-01-40 and EPA 27741-2-01-60 on 16g seats | The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Refer to Master Document List E/O No. 126-1003837 Rev. 0 (21.06.2013) or later revisions of the above listed documents approved by EASA. | As per Service Bulletin No. 126-1003837 SEB 06 00, latest revision. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair (...) | 10.07.2013 |
| 10046024 | AIRCRAFT CABIN MAINTENANCE GmbH | | | | SEE EASA APPROVED MODEL LIST | Installation of refurbished Cabin Attendant Restraint System P/NEPA2121RS-4-03-XX on 16g seats. | The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Refer to Master Document List E/O No. 127-1003944 Rev. 0 (01.07.2013) or later revisions of the above listed documents approved by EASA. | As per Service Bulletin No. 127-1003944 SEB 06 00, latest revision. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair (...) | 07.08.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|---------------------|----------------|------------------------|---|---|---|---|---|---------------|
| 10033896 | AIRCRAFT DESIGN & CERTIFICATION Ltd | | | | VENTUS CM | Steerable Tail Wheel | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Annex to the Flight Manual Ventus cM, issue 25 January 2011. Powered Sailplanes Maintenance manuals, Annex for Schempp-Hirth Ventus cM, equipped with ADXC-DC-38-001 steerable tail wheel. Anhang zum Flughandbuch Ventus cM, Ausgabe 25 Januar 2011. Motors (...) | Ventus cM with Schempp-Hirth technical Note 825-6. | 17.02.2011 |
| 10035295 | AIRCRAFT DESIGN & CERTIFICATION Ltd | | REV. 1 | | L-13 "BLANIK" | Modification and inspection program for compliance with EASA AD-2010-0185-E. | The Certification Basis of the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Modification and inspection program for compliance with EASA AD-2010-0185-E. ADXC-39-001-AFM - 30 Mar 11 - Airplane Flight Manual Supplement ADXC-39-001-AMM - 11 Jun 11 - Sailplane Maintenance Manual Supplement ADXC-39-EO-013 Issue A - 01 Apr 11 - Templ (...) | According to the applicable AFM Supplement. | 28.02.2012 |
| 10036501 | AIRCRAFT DESIGN & CERTIFICATION Ltd | | | | EA 300, EA 300/S EA 300/L, EA 300/LC | Multi glider towing capacity up to 1000kg total tow mass | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval EASA SC-023-div-02 as laid down in CRI O-09. The requirements for environmental protection (...) | Master Document List ADXC-44-001-MDL, dated August 18, 2011. Airplane Flight Manual Supplement ADXC-44-001-AFM, dated August 30, 2011 or later revisions of the above listed documents approved by EASA | Aircraft conforming to EASA TCDS A.362 and being equipped according to UA-300-4-95 Air-tow-release. | 14.09.2011 |
| 10016510 | AIRCRAFT ENGINEERING & | EASA.IM.A.S.02877 | | FAA STC ST02974AT | BOEING 767 | Installation of the KANNAD 406 ATP Emergency Locator Transmitter | | Aircraft Engineering and Installation Services Master Drawing List 234-04-0025, Revision B, 11th October 2004 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 03.06.2008 |
| 10014852 | AIRCRAFT INDUSTRIES, A.S. | EASA.A.S.03076 | | CAA CZ STC-010 | LETECKE ZAVODY L 410 UVP-E20 | Completion recorded parameters FDR BUR 1-2G for compliance with Brazilian regulation RBHA 135.152. | | Supplement No. 81 to Flight Manual of L410 UVP-E20 Supplement No. 108 to Maintenance Manual of L410 UVP-E, E9, E20 Supplement No. 108 to Wiring Manual of L410 UVP-E, E9, E20 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "#Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.03.2007 |
| 10014896 | AIRCRAFT INDUSTRIES, A.S. | EASA.A.S.03126 | | | | Implementation of VFR-NIGHT kind of operation - installation of additional equipment ensuring the extension of currently approved kinds of operation to VFR-NIGHT kind of operation. | | Supplement No. 2 to Flight Manual of Z-3FA s/n 23-08. | Applicable for s/n 23-08 only. | 23.07.2007 |
| 10027121 | AIRCRAFT PERFORMANCE MODIFICATIONS, | P-EASA.IM.A.S.03144 | REV. 1 | FAA STC SA01935LA | 401, 401A, 401B, 402, 402A, 402B, 402C, 411, 411A, 414, 414A, 421, 421A, 421B, 421C, 425, 441 | Installation of external aft body strakes | CAR 3 | - Installation in accordance with High Altitude Solutions Installation Data List, Document Number G3-050504-002, Revision C, dated November 26, 2006 or later approved Revisions - Instructions for Continued Airworthiness in accordance with High Altitude (...) | None | 22.02.2010 |
| 10015803 | AIRCRAFT SAFETY EQUIPMENT | EASA.IM.A.S.01868 | | SA3278SO | BOEING DC-10 BOEING MD 11 SERIES | Smoke Barrier Cargo Compartment McDonnell-Douglas (Boeing) MD-11F, aircraft serial no. 48445. Installation of a modified barrier in the Class "E" cargo compartment in accordance with FAA STC SA3278SO, reissued on September 24, 2004 | | 1. Aircraft Safety Equipment, Memphis, TN, Compliance Checklist, Smoke Curtain MD-11 (no date); 2. Aircraft Safety Equipment, Memphis, TN, Cargo Compartment Smoke Barrier Smoke Penetration Test Procedure 91250001, dated 31-01-1992; 3. Ship Shape Custo (...) | 1. The Aircraft Flight Manual should be amended in order to make sure that before each flight: (a) the integrity of the smoke barrier is checked; (1) the smoke barrier should be without any tears; (2) the OEM crown area smoke barrier should be without any (...) | 14.11.2006 |
| 10040586 | AIRCRAFT STRUCTURES INTERNATIONAL | | | FAA STC SA10840SC | CESSNA 208, CESSNA 208B | Installation of Jack Pads on Cessna 208 Series aircraft in accordance with FAA STC SA10840SC | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ASIC Master Data List ASIC-351 MDL, Rev. 5, dated March 23, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None | 11.07.2012 |
| 10041470 | AIRCRAFT STRUCTURES INTERNATIONAL | | | FAA STC NO. SA011054SC | 208, 208B | Installation of an internal Cargo Pod lighting system. | Design: Non-significant change. The Certification Basis of the original product and the following additional / alternative airworthiness requirements are applicable to this change: As specified in the original FAA STC No. SA011054SC dated 02-Jul-2012. Envr (...) | FAA approved installation data list document DL1919M02, rev. IR, dated 28 June 2012; FAA approved Compliance data list document DL1819M00, rev. IR, dated 28 June 2012; FAA approved Instruction for continued airworthiness 02010000-ICA rev.A, dated 8 May (...) | If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. | 19.09.2012 |
| 10043724 | AIRCRAFT STRUCTURES INTERNATIONAL | | | FAA STC SA1115SC | CESSNA 208/208B | Installation of a Standby Flap Travel Limiting System in accordance with FAA STC SA1115SC. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements / the following paragraph(s) at a later amendment: FAR 23.305(a)(b), Amdt. 45, FAR 23.1309(d)(e)(f), Amdt. 49. The requireme (...) | Master Data List ASIC208-6000-01MDL, Revision D, released September 22, 2011. Instructions for Continued Airworthiness, ASIC-002CA, Revision C, released January 6, 2012; AFM Supplement ASIC208-6000-29, Revision IR, dated December 13, 2012; or later revision (...) | The installer must determine whether this design change is compatible with previously approved modifications. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence (...) | 19.02.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|-------------------------|--|--|--|--|--|---------------|
| 10015313 | AIRCRAFT SYSTEMS AND MANUFACTURING, | EASA.IM.A.S.00166 | | FAA STC ST09887 SC | BOEING 727 | Installation of RVSM compliant equipment (dual DADS),FAA STC N° ST09887 SC | | | 1. This STC is approved only for the installation of an RVSM compliant Air Data System and does not constitute an approval of RVSM operation.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". | 07.10.2004 |
| 10015341 | AIRCRAFT SYSTEMS AND MANUFACTURING, | EASA.IM.A.S.00415 | | FAA STC ST09988SC | DC-9-81 (MD-81) DC-9-81 (MD-82) DC-9-81 (MD-83) DC-9-81 (MD-87) | Installation of Universal Avionics Terrain Awareness & Warning System. | | - FAA STC ST09988SC dated May 2002.- ASM Top Drawing List 5579-0001-00 rev. D dated 7 May 2002.- Aircraft Flight Manual Supplement No 5579-2160-00 dated 21 May 2002. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 14.12.2004 |
| 10015342 | AIRCRAFT SYSTEMS AND MANUFACTURING, | EASA.IM.A.S.00416 | | ST09989SC | DC-9-81 (MD-81) DC-9-81 (MD-82) DC-9-81 (MD-83) DC-9-81 (MD-87) | Installation of Universal Avionics Multi Function Display Model 640. | | - FAA STC ST09989SC dated 21.05.2002.- ASM Top Drawing List No 5580-0001-00 rev. D dated 07.05.2002.- AFM Supplement Doc. No 5580-2160-00 dated 21.05.2002 | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 14.12.2004 |
| 10015350 | AIRCRAFT SYSTEMS AND MANUFACTURING, | EASA.IM.A.S.00522 | | FAA STC ST09586SC REV 1 | DC-9-81 (MD-81) DC-9-81 (MD-82) DC-9-81 (MD-83) DC-9-81 (MD-87) | Installation of Universal UNS Flight Management System with UL-601 Unlink Data Link System | | -FAA STC ST09586SC Rev 1-ASM Top Drawing List No 5542-0001-00 Rev 1-AFM Supplements Doc. No 5542-2160-00 Rev B and Doc No 5542-2161-00 or later approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 19.01.2005 |
| 10015772 | AIRCRAFT SYSTEMS AND MANUFACTURING, | EASA.IM.A.S.01811 | | FAA STC ST09580SC | BOEING 767-200 BOEING 767-300 | Installation of Digital Flight Data Recorder (RFDR) Expanded Parameter Upgrade for Teledyne ARINC 717 DFDAU. | | ASM Top Drawing List 5532-0001-00. | | 20.03.2006 |
| 10015950 | AIRCRAFT SYSTEMS AND MANUFACTURING, | EASA.IM.A.S.02092 | | ST10084SC | BOEING 737-100 BOEING 737-200 BOEING 737-300 | Installation of an Innovative Solutions & Support (ISS&S) digital air data system consisting of an electric altimeter, Digital Air Data Computer (DADC), Air Data Display Unit (ADDU), and an altitude alerter in accordance with ASM Master Data List 557 (...) | | - ASM Master Data List 5572-0001-00, Revision C, Dated October 2, 2002 or later EASA approved revision.- Airplane Flight Manual Supplement No 5572-21600-00, Revision IR, dated October 17, 2002, or later EASA Approved Revision | | 15.03.2007 |
| 10015964 | AIRCRAFT SYSTEMS AND MANUFACTURING, | EASA.IM.A.S.02113 | | ST09744SC | BAE 146/AVRO 146 SERIES | Installation of a Digital Flight Data Recorder (DFDR) Expanded Parameter Sensor Upgrade on a British Aerospace Avro 146-RJ65A Aircraft | | ASM Top Drawing List 5561-0001-00, Revision D, dated July 10, 2000, or later FAA approved revision | None | 19.07.2006 |
| 10016137 | AIRCRAFT SYSTEMS AND MANUFACTURING, | EASA.IM.A.S.02333 | | FAA STC NO. ST10606SC | BOEING 767-200 BOEING 767-300 BOEING 767-400 | Installation of Mode S Upgrade, Elementary and Enhanced Surveillance Parameters | | ASM Top Drawing List No. 5675-0001-00, Rev C, 08/12/2006 | | 07.03.2008 |
| 10016276 | AIRCRAFT SYSTEMS AND MANUFACTURING, | EASA.IM.A.S.02565 | | FAA STC ST10268SC | DC-9-80 (MD-80) SERIES | FAA STC ST10268SC "Installation of Universal Avionics Terrain Awareness Warning System (TAWS)". | | Aircraft Systems & Manufacturing Top drawing List 5643-0001-00A dated 20 July 2004.AFM Supplement No. 5643-2160-00 Rev. IR dated 22 July 2004. | 1. TGL 122 complied with.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the a (...) | 10.10.2007 |
| 10031221 | AIRCRAFT SYSTEMS AND MANUFACTURING, | | | FAA STC ST09040SC | B737-300, -400, -500 | Installation of Digital Flight Data Recorder System (DFDR) using a Teledyne Digital Flight Data Acquisition Unit (DFDAU) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CRI H-01 Issue 2: Special Condition - Enhanced Airworthiness program for Aeroplane System (...) | FAA STC ST09040SC dated September 08, 2006.Aircraft Systems & Manufacturing Top Drawing List No. Q4010-000-00, Rev.R dated December 11, 2006 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | none | 03.08.2010 |
| 10016366 | AIRCRAFT TECHNICAL DEVELOPMENT Inc. | EASA.IM.A.S.02688 | | FAA STC ST02122LA | AIRBUS A320 SERIES | Installation of Aerocon Engineering Company (AEC) On-Board Airstair | | 2. FAA STC ST02122LA, issued 4th March 2009, Aerocon Engineering Co. Master Drawing List, ref. Document No. AC25-A320-1200, Rev. A, dated 21.12.07 or later revisions approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent (...)) | | 14.03.2008 |
| 10016409 | AIRCRAFT TECHNICAL DEVELOPMENT Inc. | EASA.IM.A.S.02741 | REV. 3 | FAA STC ST02112LA | A318-111, -112, -121, -122 A319-111, -112, -113, -114 A319-115, -131, -132, -133 | Installation of Cabin Interior Noise Reduction Kit/Installation of new Thermal Insulation blankets in the cabin/floor of the EASA STC EASA.IM.A.S.02741 to include A318-111/112/-121/-122 Models | CS-25 - LTI | 1. FAA STC ST02112LA amended March 27, 2009, FAA Approved Drawing List, ref. AC25-NRA319-1000, Rev. F, dated 9th January 2009 or later revisions approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision (...)) | None. | 31.03.2010 |
| 10016636 | AIRCRAFT TECHNICAL DEVELOPMENT Inc. | EASA.IM.A.S.03076 | | FAA STC ST01635 LA | BOEING 737-700 BOEING 737-800 | - Installation of Cabin Interior Noise Reduction Kit. - Validation of FAA ST01635LA. | | Aerocon Engineering Company Master Document List No. AC-BB-7370 Rev. C dated December 11, 2006 or later revisions of the above listed documents approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision). | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 20.01.2009 |
| 10016645 | AIRCRAFT TECHNICAL DEVELOPMENT Inc. | EASA.IM.A.S.03088 | | FAA STC ST01568LA | BOEING 737-700 | - Installation of Cabin Interior Noise Reduction Kit. - Validation of FAA ST01568LA. | | Aerocon Engineering Company Master Document List No. AC-BB-1000 Rev. B dated November 18, 2003 or later revisions of the above listed documents approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision). | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 20.01.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|----------------------|--|--|--|--|--|---------------|
| 10032264 | AIRCRAFT TECHNICAL DEVELOPMENT Inc. | | | FAA STC ST02317LA | A330-200 | Installation of Cabin Interior Noise Reduction Kit | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- CRI A-4301-001 'EASA Supplemental Type Certification B (...) | - FAA STC ST02317LA, dated 6th October 2010- Master Drawing List, ref. doc. No. AC25-NRA330-1000, Revision C, dated 3rd June 2010- Instruction for Continued Airworthiness, ref. doc. MMS-A330-1502 Revision NC, dated 22nd July 2009 or later revisions of t (...) | Not applicable | 21.10.2010 |
| 10034009 | AIRCRAFT TECHNICAL DEVELOPMENT Inc. | | | FAA STC ST02112LA | A320 | Installation of Cabin Interior Noise Reduction Kit/Installation of new Thermal Insulation blankets in the cabin. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval 1. CRI H-01 'Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS' The re (...) | - FAA STC ST02112LA, issued 9th January 2008- FAA Approved Drawing List, ref. AC25-NRA319-1000, Rev. H, dated 17th August 2010- Maintenance Manual Supplement, ref. MMS-A319-1380, Rev. D, dated 30th March 2008 or later revisions approved by EASA in 2009 (...) | Not applicable | 25.02.2011 |
| 10038108 | AIRCRAFT TECHNICAL DEVELOPMENT Inc. | | | FAA STC ST02382LA | CL-600-2C10 | Cabin Interior Noise Reduction Systems installation Kit in Bombardier CL-600-2C10 (Regional Jet Series 700, 701 & 702) Series Aircraft | EASA.IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | - FAA STC No. ST02382LA dated 18 November 2011- Aerocon Engineering Company Master Drawing List No AC25-NRCL600-1000 Revision B dated 29 October 2011 or later approved revision- Aerocon Engineering Company Maintenance Manual Supplement Document No. MMS (...) | None | 19.01.2012 |
| 10039327 | AIRCRAFT TECHNICAL DEVELOPMENT Inc. | | REV. 2 | FAA STC ST02361LA | A340-311, A340-312, A340-313, A340-541. | Installation of On-Board Airstair This modification installs an On-Board Airstair. This STC installed the airstairs for given configuration (aircraft model / cabin location). The Revision 1 of this STC reworded the limitation section. The Revision 2 of t (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List, Ref. 1423-7000 Revision F. Instruction for Continued Airworthiness, Service and Operational Manual, Ref. MMS-A340-1452 revision C. Lift Module Instruction for Continued Airworthiness, Service and Operational Manual, Ref. MMS-A340LIF (...) | Interaction of this STC with cabin arrangement features has not been evaluated. Should this STC be part of a cabin arrangement, the interactions between this STC and any cabin or seating arrangements must be evaluated for compliance with applicable certi (...) | 06.02.2013 |
| 10016159 | AIRCRAFT TECHNICAL SERVICE, Inc. | EASA.IM.A.S.02381 | | FAA STC SA02123LA | BEECH 200 | Installation of a nose mounting truss, bulkhead, camera, electronic backplane and missile seeker head. | | Installation in accordance with Aircraft Technical Service Inc Master Data list 3006, Revision H, dated March 12, 2008, or later approved revisions. Aircraft to be operated in accordance with FAA Approved Airplane Flight Manual Supplement ref Documen (...) | This approval is limited to Aircraft Serial Number BB-777 Only. This STC cannot be removed from the subject aircraft (BB-777) without an EASA approved modification that re-establishes its conformity to the Type design standard detailed in EASA TCDS (...) | 28.04.2008 |
| 10016800 | AIRFILM CAMERA SYSTEMS, LLC | EASA.IM.R.S.01158 | | FAA STC SR01784LA | EUROCOPTER FRANCE AS 350 SERIE EUROCOPTER FRANCE AS 355 SERIE | Installation of Down Post Utility Bracket AFDP-1 that attaches to the aircraft for attachment of a payload device | | AFDP-010 rev N/C dated 17 Feb 2005 - Master Drawing List AFDP-006 rev N/C dated 25 March 2005 - Installation Instructions Flight Manual Supplement rev N/C dated 9 May 2005 | This STC approves the installation of the Down Post Utility Bracket AFDP-1 and physical installation of payloads that meet the RFM restrictions. Any electrical installations must receive further EASA approval. | 13.07.2006 |
| 10016801 | AIRFILM CAMERA SYSTEMS, LLC | EASA.IM.R.S.01160 | | FAA STC SR01785LA | EUROCOPTER FRANCE AS 350 SERIE EUROCOPTER FRANCE AS 355 SERIE | Installation of Single Pole Utility Bracket AFSP-1 that attaches to the aircraft for attachment of a payload device. | | AFSP-010 rev N/C dated 17 Feb 2005 - Master Drawing List AFSP-006 rev N/C dated 10 May 2005 - Installation Instructions Flight Manual Supplement rev N/C dated 12 May 2005 | This STC approves the installation of the Single Pole Utility Bracket AFSP-1 and physical installation of payloads that meet the RFM restrictions. Any electrical installations must receive further EASA approval. | 13.07.2006 |
| 10016802 | AIRFILM CAMERA SYSTEMS, LLC | EASA.IM.R.S.01162 | | SR01654LA | BELL 206 BELL 407 | Installation of Model G1 Utility Bracket that attaches to the aircraft for attachment of a payload device. | | AF-G1-006 rev N/C dated 14 April 2004 - Installation Instructions AF-G3-006 rev N/C dated 14 April 2004 - Installation Instructions G1-206-005 Flight Manual Supplement dated 8 September 2004 | This STC approves the installation of the Model G1 Utility Bracket and physical installation of payloads that meet the RFM restrictions. Any electrical installations must receive further EASA approval. | 13.07.2006 |
| 10016857 | AIRFILM CAMERA SYSTEMS, LLC | EASA.IM.R.S.01233 | | FAA STC SR01058LA | EUROCOPTER FRANCE AS 350 SERIE EUROCOPTER FRANCE AS 355 SERIE | Utility Mount | | Installation of the Airfilm Camera Systems camera mount system in accordance with Master Drawing List AF200-010, Revision A, dated June 27, 2000 or AF200-1 Mount Installation Document AF200-006 Revision N/C, dated June 1, 2000 or later FAA approve (...) | | 18.04.2007 |
| 10017057 | AIRFILM CAMERA SYSTEMS, LLC | EASA.IM.R.S.01489 | | FAA STC SR02137LA | EUROCOPTER DEUTSCHLAND MBB-BK | Side step bracket with Payload Capability, in accordance with FAA STC SR02137LA. | | AF145-207: Installations Instructions. RFM Supplement to EC145 Flight Manual when modified with Airfilm Camera Step Utility Mount. | | 10.11.2008 |
| 10017058 | AIRFILM CAMERA SYSTEMS, LLC | EASA.IM.R.S.01490 | | FAA STCSR02138LA | EUROCOPTER EC 135 | Side step bracket with Payload Capability, in accordance with FAA STC SR02138LA. | | AF135-006: Installations Instructions. RFM Supplement to the EC135 Flight Manual when modified with Airfilm Camera Step Utility Mount. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 10.11.2008 |
| 10033617 | AIRFILM CAMERA SYSTEMS, LLC | | | FAA STC SR02297LA | 1. BELL 212, 412 AND 412EP 2. AB412 AND AB412EP | Nose Mount Utility Bracket | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | AF412NM-010 revision N/C dated 1 October 2009 - Master Drawing List AF412NM-006 revision IR dated 25 May 2010 - Installation Instructions AF412NM-007 revision N/C dated 25 May 2010 - Flight Manual Supplement AF412NM-008 revision N/C dated 15 April 2010 - Instructions for Continued Airworthiness. | None | 31.01.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|-------------------|----------------|----------------------|---|---|--|--|---|---------------|
| 10039428 | AIRFILM CAMERA SYSTEMS, LLC | | | FAA STC SR02377LA | 429 | Airfilm Nose Utility Mount | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List AFM-NM429-10, Rev N/C, dated August 1, 2011 - Installation Instructions AFM-NM-006, dated October 7, 2011 Instructions for Continued Airworthiness AFM-NM-008, dated August 1, 2011- Rotorcraft Flight Manual Supplement AFM-NM-005, d (...) | None | 03.05.2012 |
| 10039429 | AIRFILM CAMERA SYSTEMS, LLC | | | FAA STC SR02376LA | 429 | Airfilm Tail Utility Mount | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List AFM-AM429-10, Rev N/C, dated August 1, 2011 - Installation Instructions AFM-AM-006, dated October 7, 2011- Instructions for Continued Airworthiness AFM-NM-008, dated August 1, 2011- Rotorcraft Flight Manual Supplement AFM-AM-005, (...) | None | 03.05.2012 |
| 10015632 | AIRGLAS, INC. | EASA.IM.A.S.01586 | | FAA STC SA02244AK | CESSNA 180 CESSNA 180A CESSNA 180B CESSNA 180C CESSNA 180D CESSNA 180E CESSNA 180F CESSNA 180G CESSNA 180H CESSNA 180J CESSNA 180K CESSNA 185 CESSNA 185A CESSNA 185B CESSNA 185C CESSNA 185D CESSNA 185E CESSNA A185E CESSNA A185F | Fabrication and Installation of Airglas Inc Model LH4000 hydraulically retractable wheel skis. | | Installation in accordance with Airglas Inc Drawing No LH3600-1 original issue dated January 25, 2005 and Airglas Inc Installation, Maintenance, Service instructions and Illustrated Parts List, Airglas Inc Manual No LH4000-105, dated January 25, 2005 (...) | 1. This STC is only applicable to aircraft fitted with a Tyre size of 8.50-6, Type III. | 28.07.2006 |
| 10016958 | AIRGLAS, INC. | EASA.IM.R.S.01355 | | FAA STC SH709AL | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 D | Installation of Airglas, Inc. P/N L2700-AS350 skis on Eurocopter AS350 series helicopters. This STC is a validation of FAA STC SH709AL issued on January 29, 1981 and last amended on November 13, 2000 | | Definition, installation, and maintenance: Descriptive Data List No. L2700-AS350, Rev. C dated August 22, 2007 approved by FAA, or later FAA approved revisions. Operation: AS350B FAA approved Rotorcraft Flight Manual Supplement, dated March 11, 1982, or (...) | 1. Installation of Airglas, Inc. P/N L2700-AS350 ski kit is approved on AS350B, AS350B1, AS350B2, AS350B3, AS350BA, and AS350D helicopters which are equipped with Eurocopter standard or high skid landing gear. 2. Installation and operation of the kit o (...) | 21.09.2007 |
| 10015169 | AIRLEC AIR ESPACE | EASA.A.S.03539 | | | SA226-TB | Stretcher Installation: New stretchers installation put in the aircraft with storage place and O2 bottle space. | | Airplane Flight Manual Supplement: "Supplement Manuel de Vol SA226-T(B) - Installation caviere" Edition 1, Juillet 2008 Aircraft Maintenance Manual Supplement: "Supplement Manuel de Maintenance SA-226(B) - Installation caviere" Edition 1, Juillet 2008. | none | 06.02.2009 |
| 10017102 | AIRLIFT A/S | EASA.R.S.01072 | REV. 1 | | AS 365 N2, AS 365 N3 SA 365 C, SA 365 C1, SA 365 C2 SA 365 C3, SA 365 N, SA 365 N1 | EMS-Installation/Installation of Medical Equipment and associated attachment and supports. Note: this STC is issued for the purpose of transferring STC No. EASA.R.S.01072 of Lufttransport AS to the above identified newholder. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplements :- EMS Installation kit 555 -25-212-001 N° LT36525-032- Incubator Transport System NF02 Lateral Configuration 555-25-200-100N°LT36525-029-IVA Stretcher 555-25-201-100 N°LT 36525-030, or later revisions of the above listed d (...) | none | 23.08.2012 |
| 10017289 | AIRLIFT A/S | EASA.R.S.01539 | | | EUROCOPTER FRANCE AS 365N3 | EMS kit installation | | -Installation in accordance with Airlift AS Engineering Order EO-365-25-103 Rev 0 or later approved revisions.- Airlift AS Instruction for Continued Airworthiness ICA - 365-25-103 Rev.1 or later approved revisions.-Airlift AS Flight Manual Supplement (...) | 1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2) Th (...) | 27.05.2009 |
| 10028603 | AIRLIFT A/S | | | | AS365, EC 155 | Installation of Latitude Skynode Satcom and Tracking System | CS 29, Amendment 2, with reversion to FAR 29.561 (b)(3) at amendment 29-16 The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | MDL DO-365-23-104, Revision 0, dated 26/01/2010 Flight Manual Supplement FMS-365-23-104, Rev 0, dated 22/09/2009 | None | 28.01.2010 |
| 10029680 | AIRLIFT A/S | | | | AS365N SERIES | EMS Cabin Layout Kit DO-365-25-105 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Flight Manual Supplement, FMS-365-25-105 Revision 3 or later revisions of the above listed document approved by EASA-Instructions for Continued Airworthiness, ICA-365-25-105-01 Revision 2-Master Document List, MDL-36-25-105 Revision 3-Engineering Order, EO-365-25-105 Revision 2 | None | 12.04.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|------------------|----------------|----------------------|---|--|--|---|--|---------------|
| 10030710 | AIRLIFT A/S | P-EASA.R.S.01574 | | | EC 225, AS 332, SA 330 SERIES | Installation of Latitude SkyNode Satcom and Tracking System | | Master Document List 332-23-106, Revision 0, dated July 6th, 2009 | The SkyNode Satcom and Tracking System installed on the helicopter has to be operated in accordance with the Flight manual Supplement FMS-332-23-106 Revision 1. | 08.07.2009 |
| 10032471 | AIRLIFT A/S | | | | AS 350 B, AS 350 BA, AS 350 B1 AS 350 B2, AS 350 B3 | Installation of Geophysical Equipments in accordance with Master Document List DO-350-25-107, revision 0, dated 09.11.2010 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement FMS-350-25-107, Revision 2, dated 10.08.2010. Instructions for Continued Airworthiness ICA-350-25-107-01, Revision 2, dated 20.09.2010 | A cargo mirror allowing in-flight observation of the spectrometer and EM/ magnetometer sensors must be installed | 11.11.2010 |
| 10035077 | AIRLIFT A/S | | REV. 1 | | AS365N3 | Installation of NVIS Lighting System in accordance with Airlift MOD.DO-365-33-104. Note: Revision 1 introduces new helicopter configurations. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Airlift Flight Manual Supplement: FMS-365-33-104-01, Rev.1-Airlift Flight Manual Supplement: FMS-365-33-104-02, Rev.0 or later revisions of the above listed document approved by EASA-Airlift AS 365N3 NVIS Lighting System Installation: MOD. DO-365-33-104-02 | Refer to applicable Flight Manual Supplement (FMS) for list of equipment mandatory for Night Vision Imaging System (NVIS) certification. | 20.04.2012 |
| 10035080 | AIRLIFT A/S | P-EASA.R.S.01514 | | | AS 365N2 | Installation of NVIS Lighting System in accordance with Airlift MOD.DO-365-33-101 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Airlift Flight Manual Supplement: FMS-365-33-101-01, Rev. 0 or later revisions of the above listed document approved by EASA Airlift AS 365N2 NVIS Lighting System Installation: MOD. DO-365-33-101 Rev. 0 Airlift Instruction for Continued Airworthiness: I (...) | Refer to applicable Flight Manual Supplement (FMS) for list of equipment mandatory for Night Vision Imaging System (NVIS) certification. | 24.05.2011 |
| 10036162 | AIRLIFT A/S | | | | AS350B1 AS350B2 AS350B3 EC130B4 | Installation of Bendix/King KRA 405B Radar Altimeter. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Requirements of CS 27 at amdt. 2 are complied with by the modification. The requirement (...) | Airlift AS Master Document List MDL-130-34-100 revision 0 dtd 08.08.2011. | AS350B3 - only aircraft with Arriel 2B and Arriel 2B1 engines installed. | 17.08.2011 |
| 10041311 | AIRLIFT A/S | | | | MBB-BK 117 C-2 | NVIS certification in order to enable Night Vision Goggles operation. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI G-01, Issued dated 22/08/2012. The requirements for environmental protection and (...) | Airlift A/S Flight Manual Supplement : FMS-145-33-101-01, Rev.1, 23rd April 2012; or later revisions of the above listed document approved by EASA - Airlift A/S MBB-BK117 C-2 NVIS Lighting System Installation. DO-145-33-101 Design Order, Rev.0, 25th Au (...) | Refer to applicable Flight Manual Supplement (FMS) for list of equipment mandatory for Night Vision Imaging System (NVIS) certification. This Major Change is applicable only to helicopters delivered NVIS-friendly from Eurocopter Deutschland GmbH, as def (...) | 12.09.2012 |
| 10043743 | AIRLIFT A/S | | | | AS 350 B, BA, B1, B2, B3 | Installation of REMOVABLE EXTERNAL MOUNT Described in AIRLIFT A/S Design Order DO-350-25-115 Rev. 0 Dated 01 June 2011. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Airlift A/S Rotorcraft Flight Manual Supplement FMS-350-25-115-01 Rev.5 Dated 20 February 2013; or later revisions of the above listed documents approved by EASA. Design Order DO-350-25-115 Rev. 0 Dated 01 June 2011. Instructions for continued Airworthiness (...) | OPERATION WITH A LOAD ATTACHED TO THE REMOVABLE MOUNT MUST BE CONDUCTED IN ACCORDANCE WITH THE APPROPRIATE OPERATING RULES FOR EXTERNAL LOADS. Prior to installation of this design change it must be determined that the interrelationship between this des (...) | 21.02.2013 |
| 10044362 | AIRLIFT A/S | | | | AS 350 B, AS 350 BA AS 350 B1, AS 350 B2 AS 350 B3 | Installation of Concorde RG-350 or RG-355 Lead-Acid aircraft battery as replacement for the original Nickel-Cadmium battery. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement FMS-350-24-101-01, Rev. 1 or later revisions of the above listed documents approved by EASA. Airlift A/S Master Document List DO-350-24-101, Rev 0, date 14 February 2013. Instructions for Continued Airworthiness ICA-350-24-1 01-01, Rev. 1 | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 08.04.2013 |
| 10044769 | AIRLIFT A/S | | | | AW139 | Night Vision Image System (NVIS) Certification for AW139 helicopter | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement: FMS-139-33-101-01-01, Rev 0 dated 14 March 2013 or later revisions of the above listed documents approved by EASA - Design Order: DO-139-33-101, Rev.0 dated 3 December 2012. Instruction for Continued Airworthiness: ICA (...) | Refer to applicable Flight Manual Supplement (FMS) for list of equipment mandatory for Night Vision Imaging System (NVIS) certification. - Installation of NVIS/NVG compatible lighting does not include or imply approval for flight operations with Night V (...) | 06.05.2013 |
| 10046342 | AIRLIFT A/S | | | | AB139, AW139 | AB139/AW139 Installation of SX-5 Search Light on Short Nose configuration (DO-139-33-102). | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Aircraft Flight Manual Supplement doc. FMS-139-33-102-01 Rev. 0 or later revisions of the above listed documents approved by EASA - Master Document List doc. DO-139-33-102 Rev. 0 - Instructions for Continued Airworthiness doc. ICA-139-33-102-01 Rev. 0 | This STC is only applicable to AB/AW139 S/Ns from 31005 up to 31200 (except 31007) and from 41001 to 41200. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously i (...) | 10.09.2013 |
| 10031253 | AIRMOTIVE ENGINEERING CORPORATION | | REV. 1 | FAA STC SE10129SC | SEE EASA APPROVED MODEL LIST | Installation of rotocool, valve springs, ni-resist valve guide, exhaust valve with chrome flash on stem, and rocker shaft with thrust buttons in accordance with ECI Drawing Number AEC65314, Rev E, dated February 6, 2003, or later FAA approved revision - EASA Validation of FAA STC SE10129SC | CAR 13 / FAR 33 The Certification Basis for the original products remain applicable to this certificate. | Installation data and Instructions for Continued Airworthiness are contained in ECI Service Instruction 02-8, Rev. IR, dated February 5, 2003 (>,<,> or later revisions approved by EASA in accordance with EASA ED Decision 2004/04/CF or subsequent revisions of this decision) | none | 04.11.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------|------------------|----------------|----------------------|------------------------------|---|---|---|--|---------------|
| 10014011 | AIRPLUS MAINTENANCE GmbH | EASA.A.S.01054 | | | PIPER PA-23-250 | Installation of 1. COM/VOR/ILS/GPS-System Garmin GNS 4302. ELT Kannad 406AF | | Engineering Order 003-04 Rev 1EASA approved FMS Emergency Locator Transmitter Kannad 406AF Doc No PO0011-000406 Rev 0 or later EASA approved issuesEASA approved FMS Garmin GNS 430VHF Communications Transceiver/VOR/ILS Receiver/GPS ReceiverDoc-No. PO (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 26.07.2006 |
| 10014427 | AIRPLUS MAINTENANCE GmbH | EASA.A.S.02472 | | | CESSNA 208 | Installation of a FLIR System, Inc. Ultramedia III/HD television camera. New Pilot Operating Handbook Supplement PO-0023-250000. | | Master Document List MD-0023-000000 of 05 Feb. 2006 and test and certification reports listed in it.DOA Declaration of compliance DC-0023-250000 of Feb. 2006. | VFR; no AP, Max speed (camera operating) 115 KIAS, 2 pilots and max 4 passengers (see POH Supplement for full Limitations text). | 06.02.2006 |
| 10014848 | AIRPLUS MAINTENANCE GmbH | EASA.A.S.03070 | | | BEECH B200 SUPER KING AIR | TDR-94D-007 Mode S Transponder Dual Installation. | | Installation must be in accordance with AirPlus Master Document List MD-0109-000000 Rev. 2, dated 02.07.07 or later Approved Revisions.Operation must be in accordance with AirPlusFlight Manual Supplement for TDR94D-007, PO-0109-343300 Iss.2 dated 02 (...) | | 05.07.2007 |
| 10014849 | AIRPLUS MAINTENANCE GmbH | EASA.A.S.03071 | | | CESSNA T303 | Installation of GNS480 COM/NAV/GPS, TAW58000, Sky 497 Skywatch, KCS55A Compass System, SL30 COM/NAV, KNS3DME, GWX88 Radar, MX200 MFD and GMA 347 Audio System. This installation includes also one Digiflow and Engine monitoring system. | | Installation must be in accordance with AirPlus Master Document List MD-0121-000000 Rev. 3 dated 17.Sep.07 or later Approved Revisions.Operation must be in accordance with Flight Manual Supplement PO-0121-340100 to PO-0121-340900 Rev. 1, or later ap (...) | | 17.09.2007 |
| 10014851 | AIRPLUS MAINTENANCE GmbH | EASA.A.S.03073 | REV. 1 | | TWIN COMMANDER AIRCRAFT 690 | Installation of Garmin GTX 330 Mode S Transponder | | Flight Manual Supplement Garmin GTX 330 Mode S Transponder Doc-No: PO-0124-343300, Rev 0, dated 27/06/2007 | | 05.03.2008 |
| 10014952 | AIRPLUS MAINTENANCE GmbH | EASA.A.S.03216 | | | CESSNA S550 (CITATION II) | Installation of Bendix/King KLN 900, Shadin ADC-200 and Garmin GNS 530A. | | #1 Master Document List, Rev. 01, dated 4 September 2007 or later approved revision.# Maintenance Manual Supplement Navigation System KLN 900, MM-0175-340100, Rev. 0, or later approved revision.# Maintenance Manual Supplement COM/VOR/ILS/GPS (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 19.09.2007 |
| 10015062 | AIRPLUS MAINTENANCE GmbH | EASA.A.S.03388 | | | COMMANDER AIRCRAFT 690A | GPS Antenna Installation. | | Installation must be in accordance with AirPlus Master Document List MD-0224-000000 Rev. 1 dated 05.Feb.08 or later Approved Revisions.Maintenance must be in accordance Manual Supplement MM-0224-530512 Iss. 0 dated 05. Jan 08 or later approved revision. | | 25.03.2008 |
| 10015096 | AIRPLUS MAINTENANCE GmbH | EASA.A.S.03433 | | | BEECH B200 | Installation of SKY899 Transmitter Receiver Computer and Directional Antenna NY164 Installation. | | Installation in accordance with AirPlus Master Document List Doc-No-MD-0238-000000, Rev 1, or later approved Revisions.To be operated in accordance with AirPlus Airplane Flight Manual Supplement Doc-No. PO-0238-348990, Rev 0, or later approved revis (...) | | 14.02.2008 |
| 10015219 | AIRPLUS MAINTENANCE GmbH | EASA.A.S.03634 | | | PIPER PA-42 | Installation of the Garmin GNS530, Junction Boxes, GMA 340,EX-500 and GTX330D in Piper PA-42 Series. | | Installation must be in accordance with AirPlus Master Document List MD-0356-000001 Rev. 1 dated 29.April 2009 or later Approved Revisions.Operation must be in accordance with theFlight Manual Supplements listed in the MDL MD-0356-000001 Maintenance (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.The (...) | 04.05.2009 |
| 10015240 | AIRPLUS MAINTENANCE GmbH | EASA.A.S.03667 | | | KING AIR 200 | Installation of Dual Garmin COMM/NAV/GPS GNS430AW or GNS430Wand/or Avidyne EX500 MFD interfaced withexisting WX Radarand/or L3-Com A100S Cockpit Voice Recorderand/or Single Counter Pointer Drum Alti (...) | | Flight Manual Supplements:HPB200-AM-3160-01-01 Issue 1HPB200-AM-2310-01-01 Issue 1HPB200-AM-3130-01-01 Issue 1HPB200-AM-3470-01-01 Issue 1HPB200-AM-3423-01-01 Issue 1HPB200-AM-3430-01-01 Issue 1HPB200-AM-3450-01-01 Issue 1HPB200-AM-3400-01-01 Issue (...) | 1. Restricted to aircraft which have already approved antenna locations for the new equipment.GNS430(A)W installations restricted to aircraft which have already the followingequipment installed, completely independent from the Garmin GNS430(A)W units (...) | 11.03.2009 |
| 10028933 | AIRPLUS MAINTENANCE GmbH | P-EASA.A.S.03609 | | | PA-46-310P | Installation of Garmin GNS530W, GNS430W and GMA347. | CS-23 | Installation must be in accordance with AirPlus Master Document ListMD-0359-340630 Rev. 3 dated 19th Feb. 2010 or later Approved Revisions.Operation must be in accordance with AirPlus Flight Manual SupplementPO-0359-340347, PO-0359-340430, PO-0359-340 (...) | None | 19.02.2010 |
| 10029004 | AIRPLUS MAINTENANCE GmbH | | | | CESSNA 501, 551 | Installation of Avidyne TAS 620 and Honeywell KGP 860 | EASA CS-23The certificated noise and/ or emissions levels of the original productare unchanged and remain applicable tothis certificate/ approval | Flight Manual Supplements:PO-0673-340001, Rev. 1PO-0673-340002, Rev. 1Installation Manual and Instructions for Continued Airworthiness:EW-0673-340000, Rev. 3or later revisions of the above listed documents approved by EASA | None | 25.02.2010 |
| 10029265 | AIRPLUS MAINTENANCE GmbH | P-EASA.A.S.03811 | REV. 1 | | SEE EASA APPROVED MODEL LIST | Enabling of SVT Functionalities on Garmin G5/600 System (Revision ofEASA STC 10029265) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalCS-23 Amendment 2(not required by 21A.101)The certificated noise and/ or emissions leve (...) | Flight Manual Supplement: PO-0328-340000, Rev. 3or later revisions of the above listed documents approved by EASA-Description of Modification: DE-0328-000000, Rev. 4-Engineering Work Order: EW-0328-340000, Rev. 3 | The aircraft must have been already equipped with Garmin G5/600System(s) according to EASA STC 10029265. | 19.12.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------|------------------|----------------|----------------------|--|---|--|---|--|---------------|
| 10029716 | AIRPLUS MAINTENANCE GmbH | P-EASA.A.S.03907 | REV. 1 | | SEE APPROVED MODEL LIST | Enabling of SVT Functionalities on Garmin G5/600 Avionics System(Revision of EASA STC 10029716) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23 Amendment 02 (not required by 21A.101).The requirement for environmental protection and the associated certified noise and/or emissions levels of the original product are (...) | Flight Manual Supplement PO-0491-340000 Rev. 2 or later revisions of the above listed documents approved by EASA Engineering Work Order EW-0491-340000 Rev. 2 Description of modification: DE-0491-000000, Rev. 2 | The aircraft must have been already equipped with Garmin G5/600 System(s) according to EASA STC 10029716 | 03.12.2012 |
| 10032902 | AIRPLUS MAINTENANCE GmbH | | | | SC 01, SC 01B | Inspection Program to increase operating hours MM Modification of chapter IV Airworthiness Limitations - Implementation of Inspection Program to increase operating hours | The Certification Basis for the original product remains applicable to this certificate/ approval.FAR 23, Amdt. 1-23The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are (...) | Maintenance Manual Supplement MM-1005-000000, Rev. 1 or later revisions of the above listed documents approved by EASA | None | 07.12.2010 |
| 10034024 | AIRPLUS MAINTENANCE GmbH | | | | EASA APPROVED MODEL LIST | Dual Installation of Garmin GNS 430 (A)W/GNS 530 (A)W (TAW/S). | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.EASA CS-23, Amendment 2 (elected to comply by the applicant, not required by 21A.101)Th (...) | Engineering Work OrderEW-1126-340000, Rev.1Flight Manual Supplement:PO-1126-340000, Rev.1Maintenance Manual Supplement:.....MM-1126-340000, Rev.1 or later revisions of the above listed documents approved by EASA. | None | 28.02.2011 |
| 10034382 | AIRPLUS MAINTENANCE GmbH | | | | 1.BEECH 58, 58A, G58 2.BEECH 58P, 58PA, 58TC, 58TCA | Dual Installation of Garmin GNS430(A)W / GNS530(A)W(TAW/S) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.EASA CS-23, Amendment 2 (elected to comply by the applicant, not required by 21A.101)Th (...) | Engineering Work Order: EW-1173-000000, Rev. 1Flight Manual Supplement: PO-1173-000000, Rev. 1Maintenance Manual Supplement: MM-1173-000000, Rev. 1 or later revisions of the above listed documents approved by EASA | None | 30.03.2011 |
| 10034623 | AIRPLUS MAINTENANCE GmbH | | | | 112, 112B, 112TC, 112TCA 114, 114A, 114B, 114TC | Dual installation of Garmin GNS430(A)W/GNS530(A)W(TAW/S) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.EASA CS-23, Amendment 2 (elected to comply by the applicant, not required by 21A.101).T (...) | Engineering Work Order: EW-1162-000000, Rev. 1Flight Manual Supplement: PO-1162-000000, Rev. 1Maintenance Manual Supplement: MM-1162-000000, Rev. 1 or later revisions of the above listed documents approved by EASA | None | 15.04.2011 |
| 10035120 | AIRPLUS MAINTENANCE GmbH | P-EASA.A.S.03596 | | | DHC-8-315 | VIP Cabin Interior Modification as detailed in AIRplus Description Document DE-0348-250001, Revision 3, dated 03.05.2011 | FAR 25 as per EASA.I.M.A.191The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.EASA CS-25, Amendment 4 for changed areasThe (...) | Installation must be in accordance with AIRplus Modification Package EW-0348-000010, Revision 3, dated 31.08.2009-AIRplus Master Drawing List MD-0348-000001, Revision 9, dated 11.08.2010-V-Plane IFE System Description APDHC8-DE-4420-01-02, dated 10.08 (...) | Installation is limited to DHC-8-315, S/N 0348 | 26.05.2011 |
| 10036129 | AIRPLUS MAINTENANCE GmbH | | | | SEE EASA APPROVED MODEL LIST | Installation of Honeywell EGPWS MK VI System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.CS-23 Amendment 02The requirements for environmental protection and the associated cert (...) | Flight Manual Supplement: PO-1106-340000 or later revisions of the above listed documents approved by EASA-Engineering Work Order: EW-1106-340000-Description of Modification: DE-1106-000000-Maintenance Manual Supplement: MM-1106-000000 | The aircraft must have been already equipped with a suitable display and suitable sensors. | 12.08.2011 |
| 10037559 | AIRPLUS MAINTENANCE GmbH | | | | SEE EASA APPROVED MODEL LIST | Installation of Garmin G600 Avionics System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.CS-23 Amendment 02The requirements for environmental protection and the associated cert (...) | Flight Manual Supplement PO-0972-000000, Issue 1 or later revisions of the above listed documents approved by EASA-Aircraft Model List ML-0972-000000, Issue 2-Engineering Work Order EW-0972-340000, Issue 1-Maintenance Manual Supplement MM-0972-340001, Issue 1 | None | 06.12.2011 |
| 10038318 | AIRPLUS MAINTENANCE GmbH | | | | 500 | Installation of Garmin Transponder GTX 330D on Cessna 500 Series./Installation of Garmin Mode A/C/S Transponder GTX 330D with additional diversity antenna. | EASA TCDS I.M.A.207The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product ar (...) | 1) Master Document List: MD - 1136 - 000000 rev 4.2) Flight Manual Supplement: PO - 1136 - 3400000 rev 33) Maintenance Manual Supplement: MM - 1136 - 000000 rev 24) Weight & Balance: CA - 1136 - 000002 rev 15) Aircraft Model List: ML - 1136- 000000 r (...) | None. | 06.02.2012 |
| 10045488 | AIRPLUS MAINTENANCE GmbH | | REV. 1 | | 208, 208B | Installation of Parachute Kit on Cessna 208 and 208B, consisting of:Manual roller door, handholds, cabin benches | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: "Parachuting Activity" as defined in CRI O-08The requirements for environmental protection and the associated certified no (...) | Master Data list MD-1391-000000 Issue 8 dated 28 August 2013; -Engineering Order EW-1391-000000 issue 2 dated 12 August 2013;- To be operated in accordance with Flight Manual Supplement PO-1391-000000 issue 3 dated 13/06/2013 (for 208 model);-T (...) | All limitations defined in AFMS's.The installation of this modification by third persons is subject to written permission of the approval holder and disposal of the approved appropriate documentation.Prior to installation of this modification it must (...) | 03.09.2013 |
| 10014468 | AIRTOURING LTD | EASA.A.S.02529 | | | CESSNA 441 SERIES | Installation of Dual GTX330D, RDR2000, MX20, KN63 & KGP-560TAW/S | | Airtouring Modification AT_MOD_1085Airtouring AFM Supplement AT_FMS_1085-1 | -- | 11.08.2006 |
| 10014591 | AIRTOURING LTD | EASA.A.S.02702 | | | PIPER PA-31-350 | Installation of Avionics Upgrade | | Airtouring Modification AT_MOD_1096Airtouring AFM Supplement AT_FMS_1096 & AT_FMS_1096-1 | -- | 28.07.2006 |
| 10014717 | AIRTOURING LTD | EASA.A.S.02893 | | | SOCATA TB 10 SOCATA TB 20 SOCATA TB 200 SOCATA TB 21 SOCATA TB 9 SOCATA TB.20GT | Installation of Honeywell KGP560 Class B TAW/S | | Air Touring Modification AT_MOD_1104Flight Manual Supplement AT_FMS_1104 Issue A | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 12.12.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------------|----------------|----------------------|---|---|--|---|--|---------------|
| 10014740 | AIRTOURING LTD | EASA.A.S.02923 | | | BEECH B200 | Installation of KGP560 TAWS and Dual GTX330D Mode S | | Air Touring Mod Sheet AT_MOD_1105Compliance Record AT_CMRD_1105Flight Manual Supplements AT_FMS_1105 Rev A & AT_FMS_1105-1 Rev A or later EASA approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 05.03.2007 |
| 10014874 | AIRTOURING LTD | EASA.A.S.03099 | | | CESSNA CITATION CS 551 | Installation of Avionic Upgrade | | Air Touring Limited, Modification Sheet AT_MOD_1111.Flight Manual Supplements: AT_FMS_1111, Issue A and AT_FMS_1111-1, Issue A.Flight Manual Change Sheets: AT_FMC_1111-1, Issue A,AT_FMC_1111-2, Issue A and AT_FMC_1111-3, Issue A. | Requires prior embodiment of Gamma Support Services Minor Change GSSL-MIN-CJ551-202. | 27.11.2007 |
| 10017125 | AIRTOURING LTD | EASA.R.S.01212 | | | AGUSTA A-109 C | Installation of Dual GNS430, GTX330 and PMA700B | | Flight Manual Supplement AT_FMS_1078 Issue A | | 02.11.2005 |
| 10030747 | AIRWELD, INC. | P- EASA.IM.A.S.03228 | | | 707-300C SERIES | Vapor Cycle Air Conditioning System Installation (Validation of FAA STCSA5008NM) | | -,FAA STC SA5008NM, date reissued 10 April 2009-.,World Auxiliary Power Company (WAPCO) Drawing List Number 50000,Revision #A# dated 10 June 1990 or any later revision approved by EASAin accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision) | The limitations of FAA STC SA5008NM apply | 14.07.2009 |
| 10016989 | AIRWOLF AEROSPACE L.L.C. | EASA.IM.R.S.01395 | | FAA STC SR02491CH | ROBINSON R22 ROBINSON R44 | Airwolf Rotor Blade Protective Tape.Installation of a leading edge protective tape for Robinson R22 and R44 helicopter models according to FAA STC SR02491CHNote: The recommendation for approval is given based on a review performed in accordance with (...) | | Installation instructions:Airwolf Aerospace, LLC Installation Manual RPBT-100, Rev. A dated October 20, 2007 or later FAA approved revisions. | | 07.02.2008 |
| 10036114 | AIRWOLF AEROSPACE L.L.C. | | REV. 1 | FAA STC SR02964CH | 206A, 206B, 206L 206L-1, 206L-3, 206L-4 | TENSION-TORSION STRAP - REPLACEMENT OF LIFE-LIMITED PARTAmended Airworthiness Limitation Section of the Instruction forContinued Airworthiness AA-206-ICA | The Certification Basis (CB) for the original product remains applicableto this certificate/ approval.The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Airworthiness Limitation Section of the Instruction for ContinuedAirworthiness AA-206-ICA Rev. E, FAA Approved on 19 November 2012;or later revisions of the above listed documentsapproved by EASA in accordance with EASA ED Decision 2004/04/CF (or sub (...)) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noadverse effect upon the airworthiness ofthe product. | 28.02.2013 |
| 10016590 | AIRWOLF FILTER CORPORATION | EASA.IM.A.S.03000 | | FAA STC SA02268CH | SEE ATTACHED EASA APPROVED TYP | Installation of Airwolf Filter Corp Air Oil Separator | | Installation in accordance with Airwolf Filter Corp instructions reference AFC-W360 Rev B or later approved revisions. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness ofthe product. | 19.02.2009 |
| 10015416 | AIRWORKS | EASA.IM.A.S.01138 | | FAA STC ST01739LA | B777-200, 777-300, 777-300ER | Installation of Airworks Cockpit Door Surveillance Systems. | | -FAA STC No ST01739LA.-Airworks Index List Document Number IL23010 Revision NC dated November 10, 2004 or later approved revision. Airwork AFM SupplementDocument Number AFMS230089revision Original dated January 21, 2005 or later approved revision.-Ai (...) | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.04.2005 |
| 10015818 | AIRWORKS | EASA.IM.A.S.01889 | | FAA STC ST01576LA | 146/AVRO 146 SERIES | Installation of Airworks Cockpit Door Surveillance Systems | | FAA STC ST01576LA.AirWorks Index List Document No. IL23003 Revision A dated 15 2004 or later approved revision.AirWorksAFM Supplement Document No. AFM230017 Revision Original dated January 6 2004 or later approved revision. | | 04.01.2007 |
| 10016479 | AIRWORKS | EASA.IM.A.S.02831 | | FAA STC ST02150LA | DC-9-81, DC-9-82, DC-9-83 DC-9-87 AND BOEING MD-88 | Installation of a Cockpit Door Surveillance System (CDSS). | | As per FAA approved Air Works Master Data list, document N°MDL23019, rev. B, dated 20 June 2008. | This STC may not be installed after 7th June 2010, unless:-previously approved electrical wiring interconnection system (EWIS) instructions for onctined airworthiness (ICA) have been evaluated and determined to be adequate for this installation by F (...) | 10.07.2008 |
| 10016685 | AIRWORKS | EASA.IM.A.S.C.01060 | | FAA STC ST01443LA | BOEING 737-300/-400/-500 BOEING 737-600/-700/-800 BOEING 737-900 SERIES | Installation of AirWorks Cockpit Door Surveillance Systems | | - AirWorks Master Data List MDL230096, Revision A dated 11/11/2004 or later FAA approved Revision- AirWorks AirplaneFlight Manual Supplement Document No. AFMS230096, Revision Original dated 11/02/2005 or later FAA approved revision-FAA STC No. ST01443LA amend 8/4/2005 | 1. Limitations according to FAA STC No. ST01443LA amended8/4/2005.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/en (...) | 08.08.2005 |
| 10015355 | AIRWORTHINESS RESOURCES CORPORATION | EASA.IM.A.S.00640 | REV. 1 | TCCA STC SA00-148 | BOMBARDIER CL-415 | Installation of an Integrated Liquid Spray System (ILSS)Relocation of the cockpit control panel for the Integrated Liquid Spray System (ILSS) | | # Transport Canada STC No SA00-148 Issue 1 dated 21 December 2000.# AFM Supplement C53790 Rev. C TCCA approved on 14 May or later approved revision.# Modification Data Summary C53700 Rev. A.# Modification Data Summary C53700 Rev. C.# Service Bulletin A65-13-05 | 1. Operation in accordance with AFM Supplement C53790 Rev. B or later approved revision. AMM Supplement A65-10, orlater approved revision.2. Prior to installation of this modification the installer must determine that the interrelationship (...) | 22.10.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|----------------------------|----------------|----------------------|-----------------------------|--|---|---|--|---------------|
| 10017050 | AKV Inc. | EASA.IM.R.S.01474 | | FAA STC SR01958LA | EUROCOPTER FRANCE AS 350 B2 | Installation of AKV, Inc. Ng/Np Cycle Counter P/N 350NGLTSB2 | | Master Drawing List (MDL) No. LTS-001-B2, Rev. NC, dated September 13, 2006, or later approved revisions LTS101 Cycle Counter Installation Instructions Doc. No. 350B2-300 Rev. NC or later approved revisions LTS101 Cycle Counter Instructions for Contin (...) | AS350B2 modification law STC EASA.IM.R.S.01222 issued on 20/March 2006 is required prior to the installation of the Ng/Np Cycle Counter. The Ng/Np Cycle Counter can only be installed on daylight VFR aircraft with non-FADEC equipped engine. The Ng/Np C (...) | 29.07.2008 |
| 10028164 | AKV Inc. | EASA.IM.R.S.01555 | REV. 1 | FAA SR00631DE-D | | | | | | 31.07.2009 |
| 10040578 | ALEBCO CORPORATION A/S | | REV. 1 | | 208B | Installation of Video Transmission Unit in accordance with Master DataList PP-008-MDL-04, Issue 4, dated 11 July 2012 or later approved revision. | The Certification Basis of the original product remains applicable to this change. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation of Video Transmission Unit in accordance with Master DataList PP-008-MDL-04, issue 4, dated 11. July 2012. Operation must be in accordance with Aircraft Flight Manual Supplement PP-008-AFM-07, issue 7, dated 28. June 2012. Maintenance must be b (...) | Installer must check availability of EASA-approval for one or more referenced STC when airplane is modified per that / those STC(s). | 21.01.2013 |
| 10043003 | ALENIA AERMACCHI S.p.A. | EASA 2004-2305/ENAC CS-074 | REV. 1 | | ATR 72-201/-202/-211/-212 | ATR 72 Cargo conversion - Large cargo door (LCD). This STC converts ATR 72-201 /-202/-211 /-212 models main cabin into a cargo compartment and modifies the fuselage structure to include a large cargo door. The Revision 1 retains the certificated design (...) | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: As per ENAC CRI-A01 issue 2 dated May 2002: JAR 25 Change 15 for: 25.1, 25.2, 25.23, 25.25, 25.27, 25.31, 25.301, 25.303, 2 (...) | ATR 72 Large Cargo Door Supplemental Type Design Definition, ref: No. 54/NT/0000/N316c/020154 Issue 4, dated 4 March 2004. ATR 72 LCD Master Drawing Lists, ref: No. 54/DC/0000/N310c/020121 Issue B, dated 14 June 2002 and /040009 Issue 1, dated 29 March 2 (...) | This STC amends and replaces ENAC CS-073, dated 18 July 2002. | 04.10.2013 |
| 10043004 | ALENIA AERMACCHI S.p.A. | EASA 2004-5062/ENAC CS-095 | REV. 1 | | ATR 42-400/-500 | ATR 42 - Airborne Thematic Mapper for ATR 42MP (Multi Purpose) This STC installs on ATR 42-400 and -500 MP (multi-purpose; see ENAC STC No. CS-094) an ATM system (Airborne Thematic Mapper). The Revision 1 retains the certificated design of the ENAC STC (...) | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: JAR 25 Change 15 for: -25X20, 25.21-from 25.301 to 25.581 except 25.373, 25.445, 25.477, 25.497-from 25.601 to 25.629 exce (...) | Type Design Definition Dossier No. 8766 Issue 3 dated 26 January 2004. Type Design Definition Dossier No. 8769 Issue 2 dated 26 January 2004. ATR 42-400MP /-500MP with ATM installed Mandatory Airworthiness Limitations Supplement, ref. No. 53/NT/0000/N316 (...) | This STC can be installed only on aircraft already modified by ENAC STC No. CS-094. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change a (...) | 04.10.2013 |
| 10043005 | ALENIA AERMACCHI S.p.A. | CS-086 | REV. 1 | | ATR 42-300/-320 | ATR 42 Cargo conversion - Large cargo door (LCD). This STC converts ATR 42-300 /-320 models main cabin into a cargo compartment and modifies the fuselage structure to include a large cargo door. The Revision 1 retains the certificated design of the ENAC (...) | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: As per ENAC CRI-A01 issue 2 dated May 2002: JAR 25 Change 15 for: 25.1, 25.2, 25.23, 25.25, 25.27, 25.31, 25.301, 25.303, 2 (...) | ATR 42 Large Cargo Door Supplemental Type Design Definition, ref: No. 54/NT/0000/N316c/020412 Issue 2, dated 14 March 2003. ATR 42 LCD Master Drawing List, ref: No. 54/DC/0000/N310c/020326 Issue 3, dated 25 September 2003. ATR 42-300 /-320 CARGO LCD AFM Su (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 08.10.2013 |
| 10043006 | ALENIA AERMACCHI S.p.A. | EASA 2004-5061/ENAC CS-094 | REV. 1 | | ATR 42-400/-500 | ATR 42 - Multi-Purpose This STC modifies ATR 42-400 and -500 models to obtain a "multi-purpose" (MP) model enabling passengers transportation, cargo transportation and maritime patrol. The Revision 1 retains the certificated design of the ENAC STC No. CS-0 (...) | The Certification Basis for the original product as amended by: ENAC-RAICRI A-01 for ATR42-400/-500 MP, issue 6, dated March 2004. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original (...) | ATR 42 Multi Purpose Supplemental Type Design Definition, ref: No. 53/DO/0000/N316c/030165 Issue 1, dated 27 April 2004. ATR 42 Multi Purpose Master Drawing List, ref: 53/DC/0000/N310c/040138 Issue 1, dated 27 April 2004. ATR 42-400 /-500 AFM Supplement, (...) | 1. Type Design Definition document referred in the Associated Technical Documentation section lists the ATR modifications which must be embodied on aircraft to allow for MP conversion. 2. This STC supersedes ENAC COTS CS-033, dated 15 October 1999. Prior (...) | 04.10.2013 |
| 10043128 | ALENIA AERMACCHI S.p.A. | EASA.A.S.00455 | REV. 2 | | ATR 72-202/-212 | ATR 72 Bulk Freight Conversion (BFC) - Cargo Class E. This STC converts ATR 72-202 and -212 models main cabin into a class E cargo compartment, thus enabling bulk freight operations. The Revision 2 retains the certificated design of the STC EASA.A.S.00 (...) | The Certification Basis for the original product as amended by: the following additional or alternative airworthiness requirements: As per ENAC CRI-A01 issue 4 dated December 2004: JAR 25 Change 15 for: 25.1, 25.2, 25.23, 25.25, 25.27, 25.31, 25.301, 25.3 (...) | ATR 72 BFC Supplemental Type Design Definition, ref: No. 54/NT/0000/N316c/040148 Issue 1, dated 1st december 2004. ATR 72 BFC Master Drawing List, ref: No. 54/DC/0000/N310c/040332 Issue 1, dated 21 December 2004. ATR 72-202 /-212 CARGO BFC AFM Supplement (...) | This STC is approved only for the product configuration as defined in the approved data referred to in the paragraph "description". Compatibility with other aircraft/engine configurations shall be determined by the installer. Prior to installation of the (...) | 08.10.2013 |
| 10017181 | ALIDAUNIA S.R.L. | EASA.R.S.01304 | | | A109 E | Installation of a Ginevri Incubator Type "Baby Shuttle Intensive Care" P/N 7634B complete of the attachment fittings | | Aidaunia Flight Manual Supplement No. FMS/ALI/07/Aidaunia Modification Bulletin No. 07/ALI/2005 | Agusta A109E equipped with Agusta Kit EMS P/N 109-0326-69-103. | 14.08.2006 |
| 10017242 | ALIDAUNIA S.R.L. | EASA.R.S.01436 | | | MBB-BK 117 C-1 | MB 018/ALI/2008 "Installation of GPS/Satellite System usable to locate the helicopter position. | | FM Supplement No. FMS/ALI/018 dated 4 January 2008. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 18.07.2008 |
| 10028026 | ALIDAUNIA S.R.L. | P-EASA.R.S.01508 | | | A109S | BM No. 020/AL/2008 Installation of a "Ginevri" Incubator Type "Baby Shuttle Intensive Care" | The Certification Basis for the original product remains applicable to this certificate/ approval | RFM Supplement No. FMS/ALI/020 Rev. 0 | None | 26.11.2009 |
| 10035253 | ALIDAUNIA S.R.L. | | | | A109A II | MB 022/ALI/2010 "Installation of Alternate Static Pressure Source for the Indicator: Altimeter, Airspeed & Vertical Speed" | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement Doc. FMS/ALI/022 Rev. 0 or later revisions of the above listed documents approved by EASABT DocDR/ALI/022 Rev. 0. Maintenance Manual Supplement Doc. MMS/ALI/022 Rev. 0 | None | 15.07.2011 |
| 10043065 | ALIDAUNIA S.R.L. | EASA.R.S.01562 | | | A109 S | Sat phone system usable for transmission of voice, data and GPS position. | | BM No. 021/ALI/2008 Rev. 0. SFM FMS/ALI/021/Rev. 0 | none | 15.05.2009 |
| 10046686 | ALPINE AEROTECH LTD | | | TCCA SH97-80 | 212, 412, 412EP | Installation of Alpine Aerotech Ltd. Exhaust Ejector/Deflector Assembly | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Approval Report - Manufacture of Alpine Exhaust ejector & deflector assemblies 191M-97-713, Rev. NI (it includes Cert. Basis, CCL, Master Drawing List D191M-96-048, Rev. B) or later revisions of the above listed documents approved by EASA in accordance (...) | This approval is limited to Bell helicopter models 212, 412 and 412EP (the model 412CF is not currently EASA certified). Prior to installation of this design change it must be determined that the interrelationship between this design change and any other (...) | 11.10.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|-------------------|----------------|----------------------|---|---|--|---|---|---------------|
| 10038498 | ALPINE AVIATION, L.L.C. | | | FAA STC SA00469NY | 1- G36 2- 33, 35, 36 | Installation of Main Landing Gear Auxiliary Landing Lights in accordance with FAA STC SA00469NY/Installation of Main Landing Gear Auxiliary Landing Lights in accordance with Alpine Aviation, Inc. Installation Manual, Report No. 014, dated June 14, 1996. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Alpine Aviation, Inc. Installation Manual, Report No. 014, dated June 14, 1996/FAA approved Alpine Aviation, Inc. Airplane Flight Manual Supplement for Beechcraft Model 33 series, 35 series, 36 series with Main Landing Gear Auxiliary Landing Lights, da (...) | None. | 21.02.2012 |
| 10016940 | ALTAIR AVIONICS CORPORATION | EASA.IM.R.S.01332 | | FAA STC SR00170BO | AGUSTA AB-412 BELL 212 BELL 222 BELL 222J BELL 412 BELL 412 EP EUROCOPTER DEUTSCHLAND BK117 A EUROCOPTER DEUTSCHLAND BK117 B EUROCOPTER DEUTSCHLAND BK117 C | Monitoring System "Altair Avionics SmartCycle" | | Flight Manual Supplement FMS EMU-G-250-2 Revision B, dated 20/08/2003. | This STC supersedes EASA STC 2004-1554 by adding model Bell 412 EP to the applicability list. | 18.05.2007 |
| 10038718 | ALTITUDE AEROSPACE Inc. | | | TCCA STC SA11-43 | DHC-8-201/202 | Installation of an Oven and a Coffee Brewer in the Galley/TCCA STC SA11-43, Issue 1 | FAR 25-25 as per EASA.IM.A.191, Issue 5 (incl. SC H-01 ICA on EWIS)The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation must be in accordance with Altitude Aerospace Inc. Modification Summary MS250044, Revision E, dated 25 April 2011. Maintenance must be in accordance with Altitude Aerospace Inc. Instructions for Continued Airworthiness ICA 250044, Rev. A, (...) | Galley PN H82530009 only | 09.03.2012 |
| 10032055 | ALTITUDE AEROSPACE INTERIORS Ltd. | | REV. 1 | | B747-400 | Installation of Honeywell / Racal MSC6000 SatCom System per Altitude Aerospace Interiors Master Data List MDL 22001052 Rev. B | The Certification Basis for the original product remains applicable to this certificate/ approval. | Altitude Aerospace Instructions for Continued Airworthiness report TR35001213 Rev B or later revisions of the above listed documents approved by EASA | none | 15.11.2010 |
| 10032069 | ALTITUDE AEROSPACE INTERIORS Ltd. | | | | B747-400 | Refurbishment of the B747-400 Upper and Main deck interior installation and seating configuration in accordance with Altitude Aerospace Interiors Master Data List MDL 22001038 Rev. A or later EAWSA approved revision. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. -CRI D-01 Width of Aisle front seats Zone A-CRI D-02 Inflatable Restraint fro Head Inj (...) | Instructions for Continued Airworthiness Altitude report TR35001213 Rev. B shall be incorporated in the airplane maintenance documentation or later revisions of the above listed documents approved by EASA | This STC is applicable to A/C Serial Numbers 24855, 24896, 24957, 25605, 26910, 27602 and 29375. Several parts installed with this STC deviate from Part 21A.804 regarding EPA marking. The responsible NAA grant a 14.4 exemptions to accept the aircraft on (...) | 06.10.2010 |
| 10028359 | AMAC AEROSPACE SWITZERLAND AG | P-EASA.A.S.03791 | REV. 1 | | A320 | Humidification and Zonal Dryer System Installation (see project description document PD-AA3201-2100) | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | - Master Data List doc. No MDL-A3201-2100-1 Rev D dated 25/02/2010-Instructions for Continued Airworthiness doc. No ICA-A3201-2100 Rev C dated 24/02/2010-Declaration of Compliance Form AA-DO-77 - Rev B dated 03/03/2010 with reference to Master Data List (...) | According to doc. No ICA-A3201-2100 as listed under associated technical documentation | 04.03.2010 |
| 10028365 | AMAC AEROSPACE SWITZERLAND AG | P-EASA.A.S.03792 | REV. 1 | | A320-232 | Interior Outfitting (see project description document PD-AA01-2500-1) The limitation for use of Supplemental Water Tanks has been removed at Rev. 1 of EASA STC 10028365. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval EASA Supplemental Type Certification Basis CRI A-2301-001 Issue 2 The certificated noise (...) | - Master Data List doc. N° MDL-AA01-2500-1 Rev E dated 05/01/2010-Instructions for Continued Airworthiness doc. N° ICA-AA01-2500 Rev C dated 17/12/2009-Airplane Flight Manual Supplement doc. N° AFMS-AA01-2500-1 Rev A dated 04/12/09-Declaration of Compl (...) | Restricted to MSN 3379 | 08.01.2010 |
| 10032043 | AMAC AEROSPACE SWITZERLAND AG | | | | A319-133CJ | MSN4114 Interior Outfitting. The change consists in a complete cabin interior outfitting with a 28 passenger VIP interior including two medical stretcher units. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval Certification basis as described in CRI A-4301-003 (EASA Supplemental Type Certificatio (...) | - Master Data List Doc. N° MDL-AI01-2500 Rev B dated 04.10.2010- Flight Manual Supplement Doc. N° FMS-AI01-2500 Rev A dated 25.09.2010- Instructions for Continued Airworthiness Doc. N° ICA-AI01-2500 Rev B dated 04.10.2010 or later revisions of the above listed documents approved by EASA | MSN 4114 | 05.10.2010 |
| 10035437 | AMAC AEROSPACE SWITZERLAND AG | | | | A319-133 | Satcom and LAN / WLAN System Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. 1. CRI H-01 "Enhanced Airworthiness Programme for Airplane Systems -ICA on EWIS" The re (...) | - Master Data List, ref. doc. MDL-BL01-2300, Rev. A, dated 1st June 2011- Certification Plan, ref. doc. CP-BL01-2300, Rev. B, dated 20th June 2011- Compliance Check List, ref. doc. CCL-BL01-2300, Rev. B, dated 23rd March 2011- Declaration of Compliance (...) | Not Applicable | 22.06.2011 |
| 10036061 | AMAC AEROSPACE SWITZERLAND AG | | | | B737-700IGW | VIP seat change / Replacement of single seat by one with massage feature. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Data List: MDL-AY01-2520 rev. B dated 20.01.2011 or later revisions of the above listed documents approved by EASA. | This approval is limited to Aircraft B737-700IGW, Serial Number 36027. | 09.08.2011 |
| 10037123 | AMAC AEROSPACE SWITZERLAND AG | | | | A319-133 | Monitor Replacement / Exchange of Plasma-TV with LCD-TV screens in VIP cabin. | JAR25, Amdt. 11 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI-D 4310-009 The certificated noise and/or emissions levels of the o (...) | - Master Document List BK02-4420 Engineering Order EO-BK-4420r later revisions of the above listed documents approved by EASA. | S/N: 2421 | 04.11.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|----------------------|----------------|----------------------|---|---|--|--|---|---------------|
| 10039943 | AMAC AEROSPACE SWITZERLAND AG | | | | A319-133 | Interior Refurbishment By:- Carpet and Curtain replacement- Seat re-upholstery- Replacement of 4 double seats by 2 L-Shape Divans- VCC (IFE) Cabinet relocation- IFE upgrade- Additional Power Outlets installation- Emergency Equipment replacement | JAR 25, Chg.: 11, except as noted in TCDS EASA.A.064The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Data List, Doc. No. MDL-CF01-2500, Rev. B, dated, 01.06.2012;- AFMS, Doc. No. AFMS-CF01-2500, Rev. A, dated 01.06.2012;- ICA, Doc. No. ICA-CF01-2500, Rev. A, 31.05.2012 or later revisions of the above listed documents approved by EASA. | S/N: 1335 ,A7-HHJ | 04.06.2012 |
| 10041797 | AMAC AEROSPACE SWITZERLAND AG | | | | B777-200LR | Interior Outfitting project BA01-2500 (S/N 40753)The change consists in a complete cabin interior outfitting of B777-200LR S/N 40753 aircraft with a 125 passengers, plus 16 crew seating arrangement in a new "VIP" Cabin Interior suitable for "Private/Smoking" operation. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:As per CRI A-01 TEASA Supplemental Type Certification Basis for AMACAerospaceAG B777-20 (...) | Master Data List doc. N° MDL-BA01-2500 Rev. C dated 16.10.2012;or later revisions of the above listed documents approved by EASA. | S/N 40753 | 16.10.2012 |
| 10045537 | AMAC AEROSPACE SWITZERLAND AG | | | | 777-300ER | Interior Outfitting B777-300 ER, S/N36563 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:Special Condition(s):D-02 Isolated Compartments-D-07 Glass Screens of Displays / MonitorsD-11 Shower InstallationD-17 Inst (...) | Master Data List MDL-AV01-2500 Rev. B dtd 3. July 2013AFM supplement AFMS-AV01-2500 Rev. A dtd. 3. July 2013Instructions of Continued Airworthiness including Airworthiness Limitations ICA-AV01-2500 Rev. B dtd. 24. June 2013or later revisions of the above listed documents approved by EASA | -S/N36563 under private use-Limitations per AFM supplement AFMS-AV01-2500 Rev. A dtd. 3. July 2013-Airworthiness Limitations per Instructions of Continued Airworthiness including Airworthiness Limitations ICA-AV01-2500 Rev. B dtd. 24. June 2013 | 04.07.2013 |
| 10046155 | AMAC AEROSPACE SWITZERLAND AG | | | | A319-133 (CJ) | Interior Outfitting on A319-133As defined in the- Project Description, Doc. No. PD-BY01-2500, Rev. C, 12.08.2013- Master Data List, Doc. No. MDL-BY01-2500, Rev. C, 22.08.2013 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:Special Conditions:-CRI D-4302-015 Isolated Compartments- CRI D-4309-015 Improved Fla (...) | - Master Data List, Doc. No. MDL-BY01-2500, Rev. C, 22.08.2013- AFMS, Doc. No. AFMS-BY01-2500, Rev. E, 22.08.2013- ICA, Doc. No. ICA-BY01-2500, Rev. A, 21.06.2013 or later revisions of the above listed documents approved by EASA. | Galley G1 & G2 (Includes mechanical Installation and functional Moving of Galley Door) installed by the STC with the EASA Project Number Ref.:0010021319-001Note: Door Status Indication, Frangibility and Decompression is part of the AMAC Cabin Interior (...) | 23.08.2013 |
| 10046307 | AMAC AEROSPACE SWITZERLAND AG | | | | A340-313 | Interior Reconfiguration | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:- Special Condition(s): D-07 Glass Screens of Displays / Monitors- Deviation(s): D-06 Heat Release and Smoke DensityThe (...) | - Master Data List MDL-DF01-2500 Rev. B dtd 6. September 2013- Project Description Document PD-DF01-2500 Rev. E dtd. 5. September 2013- AFM supplement AFMS-DF01-2500 Rev. C dtd. 5. September 2013- Instructions of Continued Airworthiness ICA-DF01-2500 Re (...) | - Limitations per AFM supplement AFMS-DF01-2500 Rev. C dtd. 5. Sep 2013-The installation of this STC is limited to MSN 528- This STC does not constitute the approval of installations defined by FAA STC ref. ST10936SC dated 10 March 2009 that have not b (...) | 06.09.2013 |
| 10046671 | AMERICAN AIRLINES, Inc. | | | FAA STC ST10486SC-D | 757-200 | Interior reconfiguration, new lavatories and In Flight Entertainment Equipment upgrade | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: For applicable regulations refer to FAA STC ST10486SC-D. The requirements for environmental protection and the associated (...) | - Master Data List, American Airlines document No. F1470-MDL, Revision G, 9 October 2012- Instructions for Continued Airworthiness, American Airlines document No. F1470-ICA, Revision C, 26 March 2012- Aircraft Flight Manual Supplement, American Airlines d (...) | This STC requires an Alternative Method of Compliance (AMOC) with Airworthiness Directive (AD) 2008-10-11. The installer must obtain an AMOC with AD 2008-10-11 before installing this STC. This modification requires the floor mounted Emergency Path Light (...) | 11.07.2013 |
| 10016760 | AMERICAN EUROCOPTER CORPORATION | EASA, IM, R.S. 01089 | | FAA STC SR09281RC-D | EUROCOPTER FRANCE EC 155 B1 | American Eurocopter LLC Modification FAA STC SR09281RC-D. Installation of a Garmin Global Positioning System (GPS) - GPS500. | | AEC Drawing List 155A82-8001 revision E dated April 24 2003or later approved revision, RFM Supplement 155FMS-003 dated April 29 2003 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 15.08.2005 |
| 10016761 | AMERICAN EUROCOPTER CORPORATION | EASA, IM, R.S. 01090 | | FAA STC SR09290RC-D | EUROCOPTER FRANCE EC 155 B1 | Installation of horizontal tip position/anti-collision strobe lights | | AE Drawing List 155A82-8007 Revision B dated April 22, 2004or later EASA approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 20.07.2005 |
| 10016769 | AMERICAN EUROCOPTER CORPORATION | EASA, IM, R.S. 01100 | | FAA STC SR09282RC-D | EUROCOPTER FRANCE EC 155 B1 | American Eurocopter LLC Modification FAA STC SR09282RC-D. Installation of a Honeywell Enhanced Ground Proximity Warning System. | | ABC Drawing List 155A82-8004 revision C dated April 17 2003or later approved revision, RFM Supplement 155FMS-005 dated April 22 2003 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.08.2005 |
| 10016778 | AMERICAN EUROCOPTER CORPORATION | EASA, IM, R.S. 01117 | | FAA STC SH8804SW-D | EUROCOPTER FRANCE EC 155 B1 | Installation of a Universal Navigation UNS- 1F Flight Management System for IFR enroute & non-precision approaches. | | AEC Drawing List 365a73-9063 Revision B dated September 13 1996 or later approved revision, RFM Flight Manual Supplement (RFS 85fms-026 dated November 15 1996 or later approved revision). | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 10.10.2005 |
| 10016779 | AMERICAN EUROCOPTER CORPORATION | EASA, IM, R.S. 01118 | | FAA STC SR09295RC-D | EUROCOPTER FRANCE EC 155 B1 | Installation of navigation, communications & various electrical equipment | | American Eurocopter Drawing List 155A82-8008 revision A dated May 30 2005 or later approved revision, Rotorcraft Flight Manual Supplement dated July 8 2005 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 10.10.2005 |
| 10016834 | AMERICAN EUROCOPTER CORPORATION | EASA, IM, R.S. 01205 | | FAA STC SH7339SW-D | EUROCOPTER FRANCE AS 365N2 EUROCOPTER FRANCE AS 365N3 EUROCOPTER FRANCE SA 365N EUROCOPTER FRANCE SA 365N1 | Installation of an extended radome and FLIR 2000 infrared imaging system, in accordance with AEC Drawing List n° 365A73-9037, revision H dated 09 October 2000, or later FAA approved revision. | | Rotorcraft Flight Manual Supplement 365FMS 010, dated 12 October 2000, or later FAA approved revision. | | 28.08.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|-------------------|----------------|-------------------------|--|--|---|---|--|---------------|
| 10016835 | AMERICAN EUROCOPTER CORPORATION | EASA.IM.R.S.01206 | | FAA STC SH7338SW-D | EUROCOPTER FRANCE AS 365N2 EUROCOPTER FRANCE AS 365N3 EUROCOPTER FRANCE SA 365N1 | Installation of an AFT mounted retractable Nightsun Searchlight System, in accordance with AEC Drawing List n° 365A73-9036, revision F dated 19 March 1999, or later FAA approved revision. | | Rotorcraft Flight Manual Supplement 365FMS 009, dated 27 April 1999, or later FAA approved revision. | | 28.08.2006 |
| 10016836 | AMERICAN EUROCOPTER CORPORATION | EASA.IM.R.S.01207 | | FAA STC SH7022SW-D | EUROCOPTER FRANCE AS 365N2 EUROCOPTER FRANCE AS 365N3 EUROCOPTER FRANCE SA 365N1 | Installation of Radio Master switch, in accordance with AEC Drawing List No. 365A73-9034, revision B dated 25 April 2002, or later EASA approved revision | | -- | Compatibility of this modification with previously approved modifications must be determined by the installer. | 20.07.2006 |
| 10016837 | AMERICAN EUROCOPTER CORPORATION | EASA.IM.R.S.01208 | | SH7014SW-D | EUROCOPTER FRANCE AS 365N2 EUROCOPTER FRANCE AS 365N3 EUROCOPTER FRANCE SA 365N1 | Installation of vertical fin anti-collision light, in accordance with AEC Drawing List No. 365A73-9027, revision E dated 09 June 2003, or later EASA approved revision | | -- | Compatibility of this modification with previously approved modifications must be determined by the installer. | 20.07.2006 |
| 10016838 | AMERICAN EUROCOPTER CORPORATION | EASA.IM.R.S.01209 | | SH5499SW-D | EUROCOPTER FRANCE AS 365N2 EUROCOPTER FRANCE AS 365N3 EUROCOPTER FRANCE SA 365N1 | Installation of supplement strobe light, in accordance with AEC Drawing List No. H365-64-2D-101, revision L dated 06 June 2003, or later EASA approved revision | | -- | Compatibility of this modification with previously approved modifications must be determined by the installer. | 20.07.2006 |
| 10016839 | AMERICAN EUROCOPTER CORPORATION | EASA.IM.R.S.01210 | | SH8790SW-D | EUROCOPTER FRANCE AS 365N2 EUROCOPTER FRANCE SA 365N1 | Installation of a Bendix/King GPS KLN 90B or KLN 900 for IFR en route operations in accordance with AEC Drawing List 365A73-9060, revision C, dated 02/03/2005, or later EASA approved revision. | | AEC RFM Supplement No. 365FMS-047, original issue or later EASA approved revision. | | 07.12.2006 |
| 10016943 | AMERICAN EUROCOPTER CORPORATION | EASA.IM.R.S.01337 | | FAA STC SH8803SW-D | EUROCOPTER FRANCE AS 365N2 EUROCOPTER FRANCE SA 365N1 | Installation of combination position lights and strobe lights with supplemental anti-collision lights replacing the red, green and white position lights. | | 365A73-9062 Revision C dated 12 September 1996 or later approved revision - AEC Drawing List. | Compatibility of this modification with previously approved modifications must be determined by the installer. | 13.06.2007 |
| 10016944 | AMERICAN EUROCOPTER CORPORATION | EASA.IM.R.S.01338 | | FAA STC SH7018SW-D | EUROCOPTER FRANCE AS 365N2 EUROCOPTER FRANCE SA 365N1 | Installation of 4.0 inch Footwell in the passenger cabin of the Model SA365A73-9022. | | 365A73-9022 revision C dated 18 February 1991 or later revision - AEC Drawing List. | Compatibility of this modification with previously approved modifications must be determined by the installer. | 13.06.2007 |
| 10033895 | AMERICAN EUROCOPTER CORPORATION | | | FAA STC SR09094RC-D | EC 135 P1 (CDS) EC 135 P1 (CPDS) EC 135 P2 (CPDS) EC 135 P2+ EC 135 T1 (CDS) EC 135 T1 (CPDS) EC 135 T2 (CPDS) EC 135 T2+ | Installation of a collective activated tenth hourmeter in accordance with AEC Drawing List 135A82-8006. This EASA STC is based on FAA STC SR09094RC-D. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AEC Drawing List 135A82-8006, Revision D, dated 03/19/98 - Instructions for Continued Airworthiness 135MMS-006, dated 05 June 2007 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None. | 17.02.2011 |
| 10035015 | AMERICAN EUROCOPTER CORPORATION | | REV. 1 | FAA STC NO. SR09262RC-D | EC135 P1 (CDS), EC135 P1 (CPDS), EC135 P2 (CPDS), EC135 P2+, EC135 T1 (CDS), EC135 T1 (CPDS), EC135 T2 (CPDS), EC135 T2+ | Installation of a Belly and Fin Strobe lights in accordance with AEC Drawing List 135A82-8018. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AEC Drawing List 135A82-8018, Revision A, dated 08/20/01 - Instructions for Continued Airworthiness 135MMS-014, dated 07/31/01 or later revisions of the above listed documents. | None. | 23.05.2011 |
| 10035617 | AMERICAN EUROCOPTER CORPORATION | | | FAA STC SH8171SW-D | SA 365 N1, AS 365 N2 | Installation of Bendix /King EFIS 40 Electronic Flight Instrumentation System in accordance with AEC Drawing List 365A73-9056 (Revision B dated 10/1/93 or Bendix /King EFIS 50 Electronic Flight Instrumentation System in accordance with AEC Drawing List 365A73-9056 Revision B dated 6/28/94. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement 365FMS-017 Rev. A1, dated 06/15/2011 for EFIS 40 installation - Rotorcraft Flight Manual Supplement 365FMS-018 Rev. A1 dated 06/15/2011 for EFIS 50 installation or later revisions of the above listed documents approve (...) | none | 05.07.2011 |
| 10035996 | AMERICAN EUROCOPTER CORPORATION | | | FAA STC SR09306RC-D | MBB-BK 117-C2 | Installation of tailboom mounted aft ballast in accordance with American Eurocopter Drawing List 145A82-8006. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement 117FMS-224, Rev. A dated 17 Mar 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) - AEC Drawing List 145A82-8006, Rev. (...) | None | 03.08.2011 |
| 10039855 | AMERICAN EUROCOPTER CORPORATION | | | FAA STC SR09137RC-D | EC 135 P2, EC 135 P2+ EC 135 T1, EC 135 T2 EC 135 T2+, EC 135 P1 | Installation of an Extended Right Hand side Instrument Panel in accordance with AEC Drawing List 135A82-8020. This EASA STC is based on validation of FAA STC SR09137RC-D (Amdt. 10-May-2012) law the TIP. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AEC Drawing List 135A82-8020, No Revision, dated 07/05/01 or later revisions of the above listed documents. | none | 30.05.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|-------------------|----------------|----------------------|---|--|---|--|---|---------------|
| 10041261 | AMERICAN EUROCOPTER CORPORATION | | | FAA STC SR09104RC-D | EC135 P1, P2, P2+, T1, T2, T2+ | Installation of a pulsed engine and gearbox chip detector system in accordance with AEO Drawing List 135A82-8009. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement 135FMS-003 Rev. 3-Temporary Flight Manual Supplement "EC135A-79-001-SUP 1-AE" Revision 1 (only required if applicable) or later revisions of the above listed documents approved by EASA in accordance with EASA ED De (...) | None | 03.09.2012 |
| 10016415 | AMERICAN PROPELLER SERVICE | EASA.IM.A.S.02747 | | FAA STC SA5896NM | CESSNA 210F CESSNA 210G CESSNA 210H CESSNA 210J CESSNA 210K CESSNA 210L CESSNA 210M | Installation of a McCauley D3A34C404/80VA-0 Propeller. | | Installation in accordance with American Propeller Service, Installation Instructions AP404-1, dated June 1, 1992, or later approved revisions. Operation in accordance with American Propeller Service, Flight Manual Supplement for SA5896NM, dated January 14, 1993, or later approved revisions. | As defined in FAA STC SA5896NM1 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 02.10.2008 |
| 10014425 | AMES - AEROSPACE AND MECHANICAL | EASA.A.S.02470 | | | CESSNA 208 CESSNA 208A CESSNA 208B | Mount assembly for external side loads for Cessna 208 models | | AFM supplement AFS_M02700101_MBO_0322006 Instructions for continued airworthiness ICA_M02700101_MBO_0322006 | | 07.06.2006 |
| 10014680 | AMES - AEROSPACE AND MECHANICAL | EASA.A.S.02841 | | | CESSNA 550 | Supplemental Type Certificate for Cargo Net Installation on C550 MSN 0001-0626 | | EO M04600101 | CRI A-02; Registration: OE-GAD; S/N C550-0071 | 12.02.2007 |
| 10014681 | AMES - AEROSPACE AND MECHANICAL | EASA.A.S.02842 | | | CESSNA 551 | Supplemental Type Certificate for Cargo Net Installation on C551 MSN 0001-0549 | | EO M04600101 | CRI A-02; Registration (if applicable): OE-FAD; Serial number (if applicable): C551-0496 | 13.04.2007 |
| 10014818 | AMES - AEROSPACE AND MECHANICAL | EASA.A.S.03026 | REV. 2 | | A320 | Installation of an In-Flight Entertainment System (Live TV) on Airbus A320 Aircraft | Certification Basis as per EASA TCDS A.064 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of 1 (...) | - Engineering Order, ref. doc. No. EO_M08800106_CKA_06082010, Revision 0, dated 4th November 2010- Airplane Flight Manual Supplement, ref. doc. No. AFS_M06800105_CKA_21052010, Revision 0, dated 14th July 2010- Instruction for Continued Airworthiness, (...) | Not applicable | 05.11.2010 |
| 10015009 | AMES - AEROSPACE AND MECHANICAL | EASA.A.S.03311 | | | BEECH 400A | Activation of Garmin MX-20 Multifunction Display. | | ζ Engineering Order EO-M05200103_SAL_04062008, Rev. 1 ζ Airplane Flight Manual Supplement AFS_M05200103_SAL_27092007, Rev. 1 ζ Instructions for Continued Airworthiness ICA_M05200103_SAL_27092007 | None. | 13.08.2008 |
| 10015301 | AMES - AEROSPACE AND MECHANICAL | EASA.A.S.03862 | | | FALCON 2000 | Falcon 2000 MSN209 Interior Refurbishment. | | EO_M088000101_MBO_14042009-ICA_M088000101_MBO_14042009-EB_M088000101_MBO_14042009 | 1. Limited to A/C Falcon 2000, MSN 209.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect (...) | 11.05.2009 |
| 10027678 | AMES - AEROSPACE AND MECHANICAL | P-EASA.A.S.03803 | REV. 1 | | A330-223 | Airbus A330 - 200 B/C ISPSS Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01/Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane System (...) | Engineering Bulletin, ref. doc. No. EB_M06500111_MBO_02062009 /ABLD_24279_04, Revision 1, dated 20th October 2009- Engineering Order, ref. doc. No. EO_M06500111_MBO_04062009, Revision 0, dated 20th October 2009- Certification Program, ref. doc. No. (...) | This STC is limited to the following Serial Numbers: MSN 403, 432, 444, 454, 469, 476, 493 | 25.05.2010 |
| 10035593 | AMES - AEROSPACE AND MECHANICAL | | REV. 1 | | 737-700 AND 737-800 | B737-800/-700 Live TV System Removal / Major Change to EASASTC 10035593 to implement B737-700 as effective aircraft. AMES will use FAA approved Live TV documentation such as but not limited to: Drawings, Structural substantiation reports, ELA and W-<-&->-B for compliance demonstration. | The Certification Basis for the original product remains applicable to this certificate/ approval. No additional Airworthiness Requirements have been added and/or accepted due to the current change. The change is classified as non-significant and thereof (...) | The required documents are specified in Engineering Order EO_M08300110_MGR_09052011, Revision 2, dated 24.10.2012 or later revisions of the above listed documents approved by EASA. The modifications shall be done in accordance with the instructions defined (...) | This STC is only applicable for aircraft which have been previously modified by FAA Supplemental Type Certificate ST02887AT. | 09.11.2012 |
| 10036300 | AMES - AEROSPACE AND MECHANICAL | | | | 767-300 | B767-300 Avionics EMP Power 110VAC activation. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 Enhanced Airworthiness Programme for Airplane Systems - ICA on EWS has been a (...) | The installation shall be performed in accordance with Engineering Bulletin EB_M15400110_MGR_20042011, Revision 0. The aircraft shall be maintained in accordance with Instructions for Continued Airworthiness document ICA_M15400110_DBE_19052011, Revision (...) | EASA STC EASA.IM.A.S.01257 must be installed prior to the accomplishment of the changes of this approval, and the installation of the In-Seat components is not covered with this approval and must be approved prior to the accomplishment of the changes detailed with this approval. | 29.08.2011 |
| 10037822 | AMES - AEROSPACE AND MECHANICAL | | REV. 1 | | A330-201, -202, -203 A330-223, -243 | A330-200 Cabin reconfiguration. This STC installs mainly new B/C and Y/C seats and new In-Flight Entertainment system, according to various cabin configurations. The Revision 1 has taken into account a new cabin configuration, and reworded the limitations. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI D-4306-003: Application of heat release and smoke density requirements to seat materials. The requirements for environ (...) | Design Review Report, ref.: DRR_M14800101_MBO_1702201 Design Review Report, ref.: DRR_M14800101_MGR_21112012 Revision 01 Engineering Order, ref.: EO_M14800101_MBO_23032011 Revision 2 Aircraft Flight Manual Supplement, ref.: AFS_M14800101_MGR_06102011 Rev (...) | The following modifications, in accordance with their aircraft applicability, are prerequisites to the installation of this STC: - Luftansa Technics STC ref.: STDD33-03/01M - EASA Approval EASA.A.S.02580-AMES Minor Change ref.: M14800111.11082011 (Ap (...) | 15.02.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis (CB) for the original product | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|-------------------|----------------|-------------------------------|---|---|--|---|--|---------------|
| 10042733 | AMES - AEROSPACE AND MECHANICAL | | REV. 1 | | A330-223 | Cabin reconfiguration and IFE installation. Description - Summary : This is a Major Change to EASA STC 10042733 to include new A/C Layout into approved configurations, and consider similar IFE System, B/C and Y/C seats. Due to different Pre-Mod cabin layout (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Engineering Order EO_M17300103_RSP_23072012, rev 1 (Master document)- ICA, ref ICA_M17300103_CKA_10052012, rev 2- Aircraft Flight Manual Supplement, ref AFS_M17300103_CKA_30082012, rev 1 or later revisions of the above listed documents approved by EASA. | 1) Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 04.06.2013 |
| 10030118 | AMICALE DE VOLTIGE AERIENNE | | REV. 1 | | CAP 232 | Evolution of CAP 232 to CAP 332 SC. Revision 1: Modifications majeures applicables aux CAP23X et 332 SC | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. FAR 23. The certified noise and/ or emissions levels of the original product are unchanged (...) | Airplane Flight Manual Supplement CAP 332 SC Airplane Maintenance Manual Supplement CAP 332 SC or later revisions of the above listed documents approved by EASA | None. | 28.07.2011 |
| 10043112 | AMICALE DE VOLTIGE AERIENNE | EASA.A.S.02704 | | | CAP23], CAP23]EX, CAP232 | Fuselage frames 1,2 & 3 Reinforcement | | -SB AVA-BS-2006-01 for CAP232 aircraft- CAP232 AFMS - AVA-STC-2006-01-SB AVA-BS-2006-02 for CAP231/231EX aircraft | 1) Fuselage Frames 2 & 3 for CAP232 aircraft and Fuselage Frames 2 for CAP231 and CAP231 EX aircraft must be reinforced by Air Menuiserie only. 2. This STC is approved only for the product configuration as defined in the approved design data referred to (...) | 31.07.2006 |
| 10015819 | AM-SAFE INCORPORATED | EASA.IM.A.S.01891 | | FAA STC ST01866LA | JETSTREAM 4100 SERIES | Installation of AMSAFE Aviation Inflatable Restraint (AAIR) System on Front Row Passenger Seats of the Jetstream 4100 Series | | AMSAFE Master Data List E509050 Revision NC dated January 4 2006 or later approved revision. | Not valid for Jetstream Model 4124 | 17.05.2006 |
| 10015412 | AMSAFE, INC. | EASA.IM.A.S.01128 | | FAA STC SA01700LA. | CESSNA 172R CESSNA 172S CESSNA 177 CESSNA 206H | Installation of AMSAFE Aviation Inflatable Restraint (AAIR) System. | | -FAA STC No SA01700LA-AMSAFE Inc. Master Data List E208802 Revision NC dated November 10, 2004 or later approved revision.- AFM Supplement E508810 Revision IR dated November 24, 2004 or later approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.04.2005 |
| 10015687 | AMSAFE, INC. | EASA.IM.A.S.01672 | | SA01701LA | AVIAT HUSKY A-1 AVIAT HUSKY A-1A AVIATION AIRCRAFT A-1B | Installation of Am-Safe Aviation Inflatable Restraint System on pilot and copilot seats. Validation of FAA STC SA01701LA issued December 06, 2004 | | - AMSAFE Master Data List E508584, Revision #NC#, dated October 29, 2004 or later approved revision- Flight Manual Supplement E508590, Revision NC, dated September 9, 2004 or later approved revisions | The Electronic Module P/N 508358-409 and the Inflator Assembly P/N 508794-401 must be replaced every seven (7) years | 30.11.2005 |
| 10031010 | AMSAFE, Inc. | | | | SEE EASA APPROVED MODEL LIST | Installation of AmSafe Aviation Inflatable Restraints in Non 23.56 certified Airplanes in accordance with Model Specific Master Data List identified on AML SA02276AK, Rev J, dated 10 Dec. 2009 | FAR 23, Amdt 53 plus SC-D23.0785 Installation of inflatable seat belts in Non 23.56 certified Airplanes. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Approved Model List (AML) SA02276AK, Rev J Aircraft Flight Manual Supplement E508433 Rev NC Master data List per Aircraft Model Instructions for Continued Airworthiness E509500 Rev B Generic System Installation Instructions E509500 Rev B Operators Maint (...) | If Installation of the Shoulder Harness is not to approved Data defined in the STC a separate approval has to be obtained | 21.07.2010 |
| 10031808 | AMSAFE, Inc. | | REV. 1 | FAA STC ST02152LA | SEE EASA APPROVED MODEL LIST | AmSafe Aviation Inflatable Restraint (AAIR) System on various transport category airplanes. This revision 01 extends the applicability of AAIR to Reynard Reverb seat fitted on A330-300 airplanes, operated by Virgin Atlantic Airways. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. FAR 23.562 Special Conditions applicable to the installation of the AmSafe Aviation Inflatable (...) | AmSafe Master Data List -, E510380 Rev NC, dated May 1, 2008 for Boeing 737-., E511046 Rev A, dated December 21, 2009 for Airbus A330-, E511507 Rev NC, dated December 06, 2010 for Airbus A330-300 equipped with seat Reynard Reverb seat or later revision (...) | 1-., For the installation of approved seats modified by installation of the AmSafe Inflatable Restraint System in a specific cabin layout, compliance to 25.561 must be demonstrated for the individual seat at its specific location within the approval pro (...) | 14.03.2011 |
| 10015973 | AMTECH AERONAUTICAL LIMITED | EASA.IM.A.S.02124 | | SA02-132, ISSUE NO.3 | LOCKHEED 382G | Oil Dispersant Spraying System Installation on Lockheed 382B aircraft. | | Airplane Flight Manual Supplement FMS243901 Rev. 4 dated November 4, 2005 AFI Operating Manual OP2439 Rev. 3 dated August 2005 Instructions for Continued Airworthiness as detailed in Amtech ICA99201 Rev. 2, dated October 29, 2004. Maintenance Manual S (...) | Operation in accordance with Airplane Flight Manual Supplement FMS243901 Rev. 4 dated November 4, 2005 or later TCCA approved revision in conjunction with AFI Operating Manual OP2439 Rev. 3 dated August 2005 or later TCCA approved revision. Instruct (...) | 01.02.2007 |
| 10016638 | AMTECH AERONAUTICAL LIMITED | EASA.IM.A.S.03078 | | TCCA STC C-LSA06-378/D | LEARJET 45 | AirCell Access II Iridium Satcom TCCA LSTC C-LSA06-378/D | | As per Master Document List BAS45010021, rev. E (or later approved revisions) | 1 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 06.01.2009 |
| 10034839 | AMTECH AERONAUTICAL LIMITED | | | TCCA STC SA09-51, ISSUE NO. 1 | CL-600-2C10 CL-600-2D15 CL-600-2D24 | Installation of a Rosemount Aerospace Angle of Attack Sensor. | EASA.IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged (...) | Rosemount Aerospace Inc. (RMTAERO) Master Data List D06404350, Revision A or later approved revision (->-) Rosemount Aerospace Inc. Maintenance Manual Supplement and Instructions for Continued Airworthiness D06404348, Revision A or later approved revision (...) | None. | 05.05.2011 |
| 10039313 | AMTECH AERONAUTICAL LIMITED | | REV. 1 | TCCA STC C-LSA06-188/D | MF900 | Arflex C406-N ELT and Flight Displays FD600 Video Camera Installations. EASA validation of TCCA Limited STC No C-LSA06-188/D Issue 1 dated 23 May 2006. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Amtech Aeronautical Limited (AAL) Master Data List doc No. MDL392801, Rev. IR dated 23 May 2006 Instructions for Continued Airworthiness, doc No. 060509023, Rev. 0 dated 09 May 2006 Aircraft Flight Manual Supplement doc No. FMS392801, Rev. IR dated 09 (...) | Limited to S/N 141 | 27.04.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|-------------------|----------------|------------------------|--|--|---|---|--|---------------|
| 10044718 | AMTECH AERONAUTICAL LIMITED | | | TCCA STC C-LSA07-157/D | BD-700-1A10 | Miscellaneous Avionics and other Modifications | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. C-LSA07-157/D Issue 1 dated 09 May 2009. Bombardier Aerospace Master Document List EAS67017019 Revision NC dated 09 May 2007, or later revisions of the above listed documents approved by EASA in accordance with the Technical Imple (...) | Restricted to ASN 9003 | 30.04.2013 |
| 10015619 | ANCRA INTERNATIONAL LLC | EASA.IM.A.S.01571 | | ST01748LA | BOEING 757-200 | Installation of a Main Deck Cargo Handling System | | Ancra Interational LLC Controlled Drawing List DL83680 Revision A dated March 5, 2004 or later approved revision | | 23.12.2005 |
| 10015736 | ANCRA INTERNATIONAL LLC | EASA.IM.A.S.01760 | | ST11604LA-T | BOEING 747-400 SERIES | Installation of Main Deck Cargo Loading System | | Ancra International LLC Controlled Data List DL84400-10-2 Revision A dated April 28 2006 or later approved revision. | EASA STC EASA.IM.A.S.02040 must be installed in conjunction with this installation. | 27.06.2006 |
| 10015739 | ANCRA INTERNATIONAL LLC | EASA.IM.A.S.01763 | | ST01644LA | AIRBUS A300-600 | EASA validation of FAA STC ST01644LA Revision C to Ancra Controlled Drawing List DL83690 STC ST01644LA, cargo Loading System, Project Number SA11805LA-T | | the operating Instructions related to the cargo loading system P/N 83690-51 are described in ANCRA report 8928 | This STC is only applicable on Airbus A300-600 aircraft converted to freighter in accordance with EASA.A.S.00387/FAA STC ST01431NY. The aircraft must be compliant with the instructions for continued airworthiness (see ANCRA reports 8967 revision A and 8284 revision B, or later approved revision) | 07.04.2006 |
| 10015953 | ANCRA INTERNATIONAL LLC | EASA.IM.A.S.02095 | REV. 1 | FAA STC ST01926LA | BOEING 747-400 | Installation of Main Deck Cargo Loading System | | FAA STC ST01926LA Ancra International Controlled Data List DL84400-10-4 Revision A dated October 09 2006 or later approved revision. | EASA STC EASA.IM.A.S.02039 must be installed in conjunction with this installation. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Do (...) | 12.01.2007 |
| 10016063 | ANCRA INTERNATIONAL LLC | EASA.IM.A.S.02242 | | FAA STC ST01551LA | AEROSPATIALE ATR 42-300 AEROSPATIALE ATR 42-320 | Installation of Ancra mechanical main deck cargo landing system | | - Ancra Controlled Drawing List DL84230, Revision NC, dated June 20, 2003 | | 08.12.2006 |
| 10016114 | ANCRA INTERNATIONAL LLC | EASA.IM.A.S.02304 | | FAA STC ST01890LA | BOEING B737-400 | Installation of cargo loading system in accordance with FAA approved Ancra International LLC Controlled Drawing List DL84670-10, Revision B, dated March 22, 2006, or later FAA approved revision. | | Ancra International LLC Controlled Drawing List DL84670-10, Revision B, dated March 22, 2006, or later FAA approved revision. | 1. STC SA 2968SO (EASA.IM.A.S.02423), SA2969SO (EASA.IM.A.S.02424), and SA2970SO (EASA.IM.A.S.02425) must be installed in conjunction with this installation. 2. This STC is approved only for the product configuration as defined in the approved d (...) | 18.04.2007 |
| 10016408 | ANCRA INTERNATIONAL LLC | EASA.IM.A.S.02739 | | FAA STC ST01388LA | BOEING 747-200 BOEING 747-300 BOEING 747-400 BOEING 767-200 SERIES BOEING 767-300 SERIES BOEING 767-400 BOEING 767-400ER BOEING 777-200 BOEING 777-300 | Installation of Power Drive Unit (Cargo System). | | 747 Series Ancra Controlled Data Lists, DL92001, Revision E, 02/07/2007 DL92002, Revision E, 02/07/2007 DL92003, Revision B, 02/07/2007 Ancra Controlled Data List, DL92001, Revision E, 02/07/2007 Ancra Controlled Data List, DL92001, Revision E, 0 | | 31.03.2008 |
| 10016473 | ANCRA INTERNATIONAL LLC | EASA.IM.A.S.02819 | | FAA STC ST00680DE | BOEING B737-400 | Installation of mechanical cargo loading system Part Number 84980-10 & -20 on B737-400 aircraft | | FAA STC ST00680DE issued February 25, 2009. ANCRA Controlled Data List DL84980 Rev. B dated October 16, 2008. ANCRA Instructions for Continued Airworthiness, Doc No. 9138 Rev. B dated June 11, 2008- or later revisions of the above documents approved by E (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 02.03.2009 |
| 10028464 | ANCRA INTERNATIONAL LLC | | | FAA STC ST02279LA | B767-300 | Install Main Deck Cargo Loading System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate: Certification basis is as defined in EASA STC 10028430. The certificated noise and/ or emissions le (...) | Ancra International LLC Controlled data List DL85000, Revision D, 21/12/2009. Applicable revisions or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)* | To be installed on a/c converted by IAI Freighter Conversion (EASA STC 10028430) | 13.01.2010 |
| 10030694 | ANCRA INTERNATIONAL LLC | | | FAA STC ST02304LA | B737-300 | Installation of main deck cargo loading system. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Ancra International LLC Controlled Data List DL84990 Rev. C dated June 2, 2010. Instructions for Continued Airworthiness ICA Report 9494 Rev. B. For later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 01.07.2010 |
| 10033161 | ANCRA INTERNATIONAL LLC | | | FAA STC ST02319LA | B737-400 | Installation of main deck cargo loading system | The Certification Basis for the original product remains applicable to this certificate/ approval. | Controlled Data List DL84991 Rev. C, dated 27/09/2010. For later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | EASA validated equivalent to FAA STC ST01827LA must be installed in conjunction with this STC | 21.12.2010 |
| 10038884 | ANCRA INTERNATIONAL LLC | | | FAA STC ST02414LA | B757-200 | Installation of Main Deck Cargo Loading System. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate. For areas of change the Certification Basis is CS25 amendment 10. Requirements at amendment 10 or (...) | Ancra International LLC Controlled Data List, DL85110, Revision B, 10/10/2011, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | To be installed on a/c converted to a Combination Passenger/ Freighter by ST Aerospace Inc, EASA STC 10038822. | 21.03.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|-------------------|----------------|----------------------|--|--|--|--|--|---------------|
| 10043157 | ANCRA INTERNATIONAL LLC | EASA 2004-4159 | | FAA STC ST01536LA | 737-300SF | Installation of Ancra's Cargo Handling System on Channel Express B737-300SF aircraft | | ANCRA Drawing No. DL83734, dated 19 August 2003 Ref. JAA/04/50/27 | None. | 23.04.2004 |
| 10043713 | ANCRA INTERNATIONAL LLC | | | FAA STC ST00152LA | MD-11F | Installation of mechanical cargo loading system Main Deck-Installation of mechanical cargo loading system Part Number 81990-series in accordance with FAA approved Ancra International LLC Controlled Data List DL81990 Rev J (dated 9/21/2011). | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Ancra International LLC Controlled Data List DL81990 Rev J (dated 9/21/2011) and FAA approved 29 February 2012 Instruction for Continued Airworthiness Report 8284, Rev. B, dated 12/07/1998; Instruction for Continued Airworthiness Report 9548, Rev. B, da (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 19.02.2013 |
| 10014603 | APA AVIONES PIEZAS Y ACCESORIOS | EASA.A.S.02726 | | | CESSNA 421C | Installation GNS-430 and DME KN62A Garmin GNS-430 GPS/NAV/COMM system, Mid-Continent MD41D Annunciation Control Unit and Bendix King KN62 DME installed on Cessna 421CS/N 0272 (GPS approved for IFR and B-RNAV operations). | | Doc N° APA 018/03 "Programma de Certification", Edicion 1 Revision 0 dated 20 April 2005. Installation instructions in accordance with: Doc N° APA 018/03 "Orden Tecnica" Edicion 1 Revision 0- Garmin GNS430 Installation Manual 190-00140-02, Rev. 6- (...) | 1. Restricted to aircraft CESSNA 421C S/N 0272.2. The aircraft is approved for dual pilot B-RNAV operation, subjected to the conditions, limitations and procedures contained in Doc N° APA 018/03 "Supplement Flight Manual", Ed 2 Rev 0 | 29.06.2006 |
| 10016710 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.00619 | | FAA STC SR00470LA | AS 350 B/D/BA/B1/B2/B3 | Emergency Flotation Kit installation according to FAA STC No. SR00470LA issued on November 17, 1997 and reissued December 22, 1998. | | - EASA-approved AS 350 Flight Manual and RFM Supplement Apical FMS-350(2) dated October 15, 1997, or later FAA-approved revisions are required. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 14.04.2005 |
| 10016719 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01013 | | FAA STC SR00780LA | EC 120 B | Installation of Apical Emergency Float Kit, p/n 20498. The present STC is the validation of the FAA STC n° SR00780LA issued on 19/03/1999. | | - FAA STC No SR00780LA issued on 19/03/1999. Definition: Master Document List MD120-1, revision D dated March 18, 1999 approved by FAA or later FAA-approved revisions. - Installation: Installation and Packing Instructions II120-100, revision C dated March (...) | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.04.2005 |
| 10016772 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01107 | REV. 1 | FAA STC SR01779LA | AB412, AB412EP BELL 212, 412, 412EP/CF BELL 412 EP | STC EASA.IM.R.S.01107 applicability extended to B212, 412 CF, AB412 and AB412EP. | | 1. RFMS412-41, Rev. A, 11/25/05, Rotorcraft Flight Manual Supplement, MD412-41, Rev. F, 09/20/05, Master Document List, II412-41, Rev. D, 09/07/05, Installation Instructions Bell 412 Emergency Float Kit, ICA412-41, Rev. B, 07/12/05 Instr (...) | 1. Instruction for Continued Airworthiness ICA 412-41 Rev. B dated 12th July 2005 or later approved revision. 2. Bell Helicopter Textron, Inc. High Skid Landing Gear with Lightweight Flotation Skid Tubes, Part Number 412-050-015-111/-112 or Aer (...) | 14.05.2007 |
| 10016775 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01114 | | FAA STC SR00831LA | AS 350 D/B/B1/B2/B3/BA, AS 355 E/F/F1/F2/N | Installation of Apical Emergency Float kit, Part Number 20416, in accordance with the following FAA approved Apical Industries, Inc., Master Document List and Rotorcraft Flight Manual Supplement : For AS 350 series : MDL n° MD350(4) rev. I, dated (...) | | - Apical AS 350 RFMS n° FMS-350(4), rev. B dated 15/11/05 or later EASA approved revision. - Apical AS 355 RFMS n° FMS-355(2) rev. B dated 15/11/05 or later EASA approved revision. - Apical Installation and Packing Instructions n° II355-100 rev. E, dated (...) | 1. This approval requires the inspection and overhaul schedules to be performed as mandated by the Apical Instructions for Continued Airworthiness n° ICA 355-1, rev. G, dated December 3rd, 2004 or later FAA approved revision. This installation sho (...) | 15.11.2005 |
| 10016776 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01115 | | FAA STC SR01687LA | EC 130 B4 | Installation of Apical Emergency Float kit p/n 604-0018-100 or Emergency Float kit with Liferaft p/n 604-0018-300. | | Definition: Apical Master Document List n° MD130-1, revision J, dated 04/28/05, or later FAA approved revisions. Installation: Apical Installation Instructions for EC 130 B4 Emergency Float kit n° 11130-1, revision D, dated 12/22/04, or later revisions Ap (...) | This STC is the EASA validation of FAA STC n° SR01687LA, dated March 22nd, 2005. | 25.04.2008 |
| 10016788 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01135 | | FAA STC SR01535LA | BELL 206A, 206B, 206L, 206L-1 BELL 206L-3, 206L-4, BELL 407 | Installation of Apical Industries Inc. Emergency Float Kits. | | - MDPH-1 # Master Document List, FMSPHI-1, revision B dated May 13, 2005 # Rotorcraft Flight Manual Supplement, ICAPHI-1, revision G dated 15 April, 2005 # Instructions for Continued Airworthiness, ICAPHI-2, revision A dated 15 April, (...) | 1. This installation of Apical Industries Inc. Emergency Float Kit requires prior installation of high landing gear with provision for float installation as defined in Apical Industries Inc. Approved Landing Gears for Apical Emergency Float Kit, A (...) | 16.01.2007 |
| 10016795 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01148 | | FAA STC SR00627LA | AS 350 B, AS 350 B1, AS 350 B2 AS 350 B3, AS 350 BA, AS 350 D | Installation of Utility floats in accordance with FAA STC SR00627LA issued May 08, 1998, reissued December 22, 1998 and amended September 14, 1998. | | Installation in accordance with FAA approved Apical Industries Inc Master Document List MD 350(3), Revision H, dated August 11, 1998 or later FAA approved revisions. FAA approved Apical Industries Inc Rotorcraft Flight Manual Supplement FMS-350(3) dated May 05, 1997 or later FAA approved revision. | This approval requires the inspections and overhaul schedules to be performed as mandated by the Instruction for Continued Airworthiness, ICA350-2 dated 2, 1997 or later FAA approved revision. | 20.06.2006 |
| 10016829 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01199 | REV. 1 | FAA STC SR01855LA | EC 135 P1 (CDS) EC 135 P1 (CPDS) EC 135 P2 (CPDS), EC 135 P2+ EC 135 T1 (CDS) EC 135 T1 (CPDS) EC 135 T2 (CPDS), EC 135 T2+ | Emergency Floats Kits Installation of Apical Industries, Inc. Emergency Floats Kits i.a.w. FAA STC SR01855LA Apical MDL MD 135-1 provides for different installations either with liferafts or without. Amendment of the STC in order to add the EC135 Mid Hel (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | For installation i.a.w. MD 135-1 Rev. A, Rotorcraft Flight Manual Supplement RFMS 135-1, Rev. Cor later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | 1. The installation is not approved for helicopters equipped with high skid landing gear. 2. Metro Aviation, Inc., Landing Gear Skid Extensions, Part Number 135M-110, must be installed i.a.w. STC EASA.IM.R.S.01377 (FAA STCSR09230RC) prior to the instal (...) | 01.06.2011 |
| 10016831 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01202 | | FAA STC SR01459LA | BELL 412, 412EP | Installation of Emergency Mid-Float with 10-man Liferaft Kits Installation of Apical EMERGENDY Mid-Float with 10-man Liferaft Kits | | Master Document List MD-412-31, Rev K, dated 3 February 2006 or later approved revision Instruction for continued Airworthiness ICA-412-31 Rev E, dated 13 April 2005 or later approved revision Rotorcraft Flight Manual Supplement FMS-412-31 Rev D, dated 9 February 2006 or later approved revision | Bell Helicopter Textron, Inc. Emergency Float Kit Part Number 412-706-004-137, or 412-706-004-143, or 412-706-004-149 or Apical Industries, Inc. Emergency Float Kit, Part Number 604.9601 per FAA STCSR01648LA, when approved by EASA must be installed (...) | 04.09.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|-------------------------|----------------|----------------------|---|---|---------------------|--|--|---------------|
| 10016849 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01225 | | FAA STC SR00856LA | BO105 A/C/S | Installation of Apical Emergency Float Kits, Part Number 20430Description: Installation of Emergency Float Kit in accordance with FAA STC SR00856LA. Apical MDL MD105-2 provides for two different installations either with liferafts or without.Note: the (...) | | For installation i.a.w. MD105-2 Rev. B/8-18-1999;Instructions for continued airworthiness: ICA 105-2, Rev A Flight Manual Supplement FMS-105(2) Rev. AFor installation i.a.w. MDL MD 105-2 Rev. F12-08-2004;Instructions for continued airworthiness: ICA105-501, Rev. N/CFMS-105-501 Rev. A | The installation is not approved for helicopters equipped with high skid landing gear. | 07.03.2008 |
| 10016850 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01226 | | FAA STC SR00932LA | MD 369 HS, HM, HE, D,E,FF,500N | Installation of Apical Emergency Roat kit p/n 20491 -1. | | Definition:Apical Master Document Listnù MD500-10, revisionH, dated 10/12/05, or later FAA approved revisions.Installation:Apical Installation Instructions n011500-10, revision D, dated 08/03/05, or later revisionsOperation:Apical Flight Manual supp (...) | This STC is the EASA validation of FAA STC ne SR00932LA dated February 23rd, 2006. | 25.04.2008 |
| 10016851 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01227 | | FAA STC SR01420LA | AS 332 C/L/L1 | Installation of Apical Industries, Inc. Emergency float assemblies, P/N 20729-100, 207300-100 and 2073, and inflation cylinders, P/N 20330-2 | | FAA approved Apical Industries, Inc. Master Drawing List MDL332-1, rev. J dated September 18, 200, or later FAA approved revision.FAA approved Apical Industries Inc. Instructions for Continued Airworthiness ICA332-1, rev A dated January 31, 2006 or later FAA approved revision. | 1 Aerazur's emergency float assemblies P/N 158820, 158565 and 158566, and inflation cylinders P/N 158562 (i.e. parts of original emergency floatation kit approved on Eurocopter AS332C, L and L1 models) must be installed prior to applying this STC.The (...) | 13.05.2008 |
| 10016865 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01241 REV 1 | REV. 1 | FAA STC SR01902LA | S-76 A, B, C | Installation of Apical Industries Inc. Liferaft Kit. | | MD76-2, revision E dated 19 May, 2006 # Master Document ListRFMS-76, revision N/C dated 28 June, 2006 # Rotorcraft Flight Manual SupplementICA76-2, revision A dated 12 June, 2006 # Instructions for Continued Airworthiness | This STC is approved only for the product configuration as defined inthe approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation".Compatibility with other aircraft/engine configurations shallbedetermin (...) | 16.05.2007 |
| 10016923 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01308 | | FAA STC SR01949LA | AGUSTA AB-139, AW139 | Installation of Apical Industries, Inc. Liferaft Kit. | | RFMS APICAL AW139 Liferaft Kit original issue, rev. A, dated 12-10-2007, or later FAA approved revisions.Master Document List MD139-1, Rev. B, dated 21-12-2006, or later FAA approved revisions.Instructions for Continued Airworthiness ICA,139-1, rev (...) | 1. This approval requires the inspections and overhaul schedules to be performed as mandated by the Instruction for Continued Airworthiness, ICA139-1, Revision N/C, dated July 27,2006, or later approved revision. | 19.10.2007 |
| 10016941 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01334 | | FAA STC SR02049LA | ROBINSON R44, R44 II | Pop Out Floats Kit: Installation of Apical Industries, Inc.Pop Out Floats Kit (also titled as "Emergency Floats Kit") in accordance with approved Apical Industries, Inc. Master Document List MD44-1, rev. D, dated 12-04-2007, or later approved revision based on FAA STC SR02049LA. | | Flight Manual Supplement RFMS R44-2, Rev. N/C, FAA approved19-10-2007 or later approved revision.Installation Instructions: #44-1, Rev. B, dated 26-03-2007 or later approved revision.Instructions for Continued Airworthiness ICA44-1, Rev. A., dated 01-05-2007 or later approved revision. | N/A | 27.11.2007 |
| 10016947 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01341 | | FAA STC SH7685SW | BO-105 C/S | Replacement of the MBB Oil Cooler Fairing, MBB P/N 105-61802 | | FAA STC SH 7685SW Installation in accordance with ERA Drawing 10579-100-001, dated September 27, 1989, or later approvedrevision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 15.08.2007 |
| 10016971 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01372 | | FAA STC SR02065LA | AS 350 B/B1/B2/B3/BA/D AS 355 E/F/F1/F2/N | 350-31 Lower Front Fairing Assembly. | | Instructions for Continued Airworthiness, ICA350-31, Revision B, dated 3rd January 2007, or later approved revision. | 1. This approval requires the inspections and overhaul schedules to be performed as defined by Apical Industries Inc's Instructions for Continued Airworthiness, ICA350-31, Revision B, dated 3rd January 2007, or later approved revision.2. Prior (...) | 23.01.2009 |
| 10017017 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01435 | | FAA STC SR02108LA | AB139, AW139 | Nose Landing Gear Doors Kit on Agusta AB139, AW139. | | Master Document List MD139-2, Revision B, dated November 8,2007, or later approved revision.Instructions for ContinuedAirworthiness, ICA139-2, Revision A, dated August 6, 2008, or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 29.01.2009 |
| 10017022 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01440 | | FAA STC SR01439LA | BELL 407 | Emergency Float Assemblies. | | MD407-1 Revision C dated 5 February 2003 or later approved revisions # Master Document List;ICA407-1 Revision N/C dated12 June 2002 or later approved revision # Instructions for Continued Airworthiness | Bell Helicopter Textron Canada Limited Emergency Float Kit P/N 407-350-300-101 and/or 407-350-301-101 and/or 407-350-302-101 must be installed prior to the installation of this STC.The intermixed installation configuration between the following Apic (...) | 13.03.2008 |
| 10017023 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01441 | | FAA STC SR01392LA | A, AW119 MKII119 | Emergency Float Kits FAA STC SR01392LA dated 21 December 2007. | | Master Document List MD A119-1 Rev. T dated 19 November 2007 or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 12.06.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|-------------------------|----------------|-----------------------|--|---|---|--|--|---------------|
| 10017081 | APICAL INDUSTRIES, Inc. | EASA.IM.R.S.01534 | | FAA STC SR02180LA | AS 350 B, AS 350 B1, AS 350 B2 AS 350 B3, AS 350 BA, AS 350 D | Installation of a Cable Cutter Kit. | | # Apical Master Document List MD350-61, Rev. A dated November 25, 2008, later EASA approved revision# Instructions for Continued Airworthiness for AS350 Cable Cutters Kit P/N 646.4001 & AS350 Wiper Deflector Kit P/N 646.4002 for Eurocopter Mode (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 23.03.2009 |
| 10028171 | APICAL INDUSTRIES, Inc. | | | FAA STC SR02244LA | BK117 C-2 | Title: Emergency Float KitsDescription: Installation of Apical Industries, Inc. Emergency FloatKits. This STC is based on the validation of FAA STC SR02244LA issued21.07.2009. | FAR 29 as per EASA TCDS R.010The Certification Basis for the original product remains applicable tothis certificate/ approvalThe certificated noise and/ or emissions levels of theoriginal productare unchanged and remain applicable to thiscertificate/ approval | - Flight Manual Supplement RFMS145-1, Rev. N/C, approved July 20, 2009or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions ofthis decision)- Master Document List MD1 (...) | None | 07.12.2009 |
| 10028493 | APICAL INDUSTRIES, Inc. | | | FAA STC SR00645LA | AS355E, AS355F, AS355F1 AS355F2, AS355N, AS355NP | Emergency Float Kit | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/approval. | MD355(1) revision F dated 1 July 1998 - Master Document List;FMS-355(1) dated 29 July 1998 - Flight Manual Supplement;CA350-1 revision C dated 12 January 1998 - Instructions for ContinuedAirworthiness;CA350-20 revision B dated 3 December2004 - Ins (...) | Not Applicable | 16.03.2011 |
| 10029348 | APICAL INDUSTRIES, Inc. | | | FAA STC SR02271LA | 206A, 206B, 206L, 206L-1, 206L-3, 206L-4 | Bell 206 Cable Cutter Kit | FAR 27The Certification Basis for the original product remains applicable tothis certificate/ approvalThe certificated noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval | MD206-21, revision A dated 10 November 2009 - Master Document List;-ICA206-21 revision N/C dated 16 July 2009 - Instructions for ContinuedAirworthiness.or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Deci (...) | None | 19.03.2010 |
| 10030503 | APICAL INDUSTRIES, Inc. | | | FAA STC SR02289LA | BELL 204B BELL 212, BELL 214B BELL 214B-1, BELL 205A-1 BELL 412, BELL 412EP | Installation of Cable Cutters Kit in accordance with Master DocumentList MD412-101, Revision B, dated May 12, 2010 or later FAA approved.revision. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval (*)FAR 29 per Apical PSCP412-101, Rev. C | Instructions for Continued Airworthiness ICA412-101, Revision N/C, datedJanuary 5, 2010or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions ofthis decision) | None | 21.06.2010 |
| 10031793 | APICAL INDUSTRIES, Inc. | | | FAA STC SR02307LA | EC135 P1(CDS), EC135 P1(CPDS) EC135 P2(CPDS), EC135 P2+ EC135 T1(CDS), EC135 T1(CPDS) EC135 T2(CPDS), EC135 T2+ | Cable Cutters KitInstallation of Cable Cutters Kit with defectors on skids and wiper inaccordance with Master DocumentList MD135-2, based on FAA STCSR02307LA. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | - Master Document List MD135-2, Rev. A, dated June 24, 2010- Instructions for Continued Airworthiness ICA135-21, Rev. N/C, datedJanuary 13, 2010or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004 (...) | none | 14.09.2010 |
| 10032468 | APICAL INDUSTRIES, Inc. | | REV. 1 | FAA STC SR02290LA | MBB-BK 117 C-2 | Cable Cutters KitInstallation of Cable Cutters Kit with defectors on skids and wiper inaccordance with Master DocumentList MD145-2, based on FAA STCSR02290LA. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval | - Master Document List MD145-2, Rev. A, dated August 16, 2010- Instructions for Continued Airworthiness ICA145-21, Rev.N/C, datedSeptember 23, 2010or later revisions of the abovelisted documents approved by EASA inaccordance with EASA EDDecision 2004/04/CF (or subsequent revisions ofthis decision) | None | 19.11.2010 |
| 10034031 | APICAL INDUSTRIES, Inc. | P- EASA.IM.R.S.01426 | | FAA STC SR00471LA | MD600N | Emergency Floats | The Certification Basis for the original product remains applicable tothis certificate/ approvalThe requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval | Master Document List MD600-1, Revision B, dated September 30, 1997,plus:- Report A1600N-3, Rev A, dated 03/31/09- Report A1600N-1, Rev D, dated 03/31/09- Rotorcraft Flight manual Supplement FMS-600(1) Rev C Instructions for Continued Airworthiness (...) | None | 01.03.2011 |
| 10034100 | APICAL INDUSTRIES, Inc. | | REV. 1 | FAA STC SR02291LA | BELL 407 | Cable Cutters KitThe Bell 407 Cable Cutter Kit provides a means of protecting thehelicopter during a wire strike by providing the capability to cut acable or wire before catastrophic damage occurs. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | - Master Document List MD407-41, Revision A, dated 23 November 2010;- Instructions for Continued Airworthiness ICA407-41, Revision N/C,dated 8 February 2010;- Installation Instructions I407-41 revision A dated 22 November 2010.or later revisions to the above. | None | 21.03.2011 |
| 10039222 | APICAL INDUSTRIES, Inc. | | | FAA STC SR02379LA | AW139 | EASA Validation of FAA STC No. SR02379LA "Installation of ApicalIndustries, Inc. Cable Cutter Kit" | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/approval. | Apical Industries, Inc. Master Document List MD139-3 Rev. B, dated 17October 2011Apical Industries, Inc. Instructions for Continued Airworthiness, ICA139-31, Rev. N/C, dated 15 July 2012 | None | 18.04.2012 |
| 10041959 | APICAL INDUSTRIES, Inc. | | | FAA STC NO. SR02425LA | A119, A119 MK II | EASA Validation of FAA STC No. SR02425LA "Installation of ApicalIndustries, Inc. Cable Cutter Kit". | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Apical Industries, Inc. Master Document List MD119-2 Rev. B, dated 17April 2012;Apical Industries, Inc. Instructions for Continued Airworthiness, ICA119-2, Rev. A, dated 19 April 2012;or later revisions of the above listed documents approved by EASA (...) | None | 26.10.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------|-------------------|----------------|----------------------|--|---|--|--|---|---------------|
| 10042766 | APICAL INDUSTRIES, Inc. | | | FAA STC SR02373LA | EC130B4 | Installation of Fresh Air Vents in accordance with Apical Industries, Inc. Master Document List MD130-3. This STC is the validation of FAA STC SR02373LA (date of issuance September 12, 2012). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Apical Industries, Inc. Master Document List MD130 3 Rev. Fdtd17/07/2012; Apical Industries, Inc. Rotorcraft Flight Manual Supplement RFMS130 3 Rev. N/C dtd 03/08/2012; Apical Industries, Inc. Instructions for Continued Airworthiness ICA1303 Rev. B dt (...) | None. | 18.12.2012 |
| 10016824 | APOLLO SPRAY SYSTEMS | EASA.IM.R.S.01194 | | SR01215CH | ROBINSON R44 | Installation of Apollo Spray system DTM-4 agricultural sprayer onto Robinson R44 helicopter. EASA validation of FAA STCSR01215CH | | Apollo spray systems Document DTM-4-510 .FAA approved Rotorcraft flight manual supplement (Metric version), dated 21 April 2006, or later FAA approved revisions Apollo Spray system Drawing List DTM-4-MDL. No rev. dated December 8, 1999, or later FAA a (...) | Certified in restricted category. All limitations of RFM supplement apply. | 12.02.2007 |
| 10042428 | APPAREO SYSTEMS, L.L.C. | | | FAA STC SR02797CH | AS350B1, AS350B2, AS350B3, AS350BA. | Installation of Vision 1000 System in accordance with Vision 1000 Master Document List 606586-000011 This STC is the validation of FAA STC SR02797CH (date of issuance December 28, 2009) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Appareo Systems Vision 1000 Master Document List 606586-000011, Revision 1.10 dtd 27 April, 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 30.11.2012 |
| 10015635 | ARC AVIONICS CORP. | EASA.IM.A.S.01591 | | FAA STC ST02959AT | BOEING 737-300 SERIES BOEING 737-400 SERIES BOEING 737-500 SERIES | Installation of Collins Model 700 High Frequency (HF) Communication System and HF Antenna | | 1. ARC Avionics Master Drawing List, Ref. ARC-HF, 001-MDL, Rev. B dated 17 September 2004 or later FAA approved revision 2. ARC Avionics Airplane Flight Manual Supplement, Ref. ARC-HF.001-FMS, Rev. IR, dated 22 October 2004 or later FAA approved revision 3. FAA STC ST02959AT, dated 22 October 2004 | According to FAA STC ST02959AT | 17.11.2005 |
| 10015665 | ARC AVIONICS CORP. | EASA.IM.A.S.01641 | | FAA STC SA2890AT | FAIRCHILD SWEARINGEN SA 226 FAIRCHILD SWEARINGEN SA 227 FAIRCHILD SWEARINGEN SA 26 | Installation of a Honeywell Model CAS 67A Traffic Alert and Collision Avoidance System II (TCAS II) with Single Mode-S Transponder MST 67A iaw FAA STC SA02890AT. | | 1. ARC-TCAS 001-GAW (Continued Airworthiness) 2. ARC-TCAS.001-FMS (AFM Supplement) 3. FMS ASR 2006-041-05, Rev. A, COM 2 GND Clearance Function. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 05.09.2008 |
| 10016083 | ARC AVIONICS CORP. | EASA.IM.A.S.02267 | | FAA STC03354AT | BOEING 737-200 BOEING 737-300 | Installation of a single Universal Avionics Class #A# TAWS system on 737-300 series aircraft. | | - ARC Avionics Master Data List 760-061505, Rev E, dated November 10, 2006, or later approved revision. - Flight Manual Supplement Doc 730-061505 Rev A, dated Feb 08 2007. | The aircraft has to be equipped with a Universal EFIS to display the TAWS terrain. EASA Approved Flight Manual Supplement ARC Avionics document 730-061505 Rev A for Boeing 737-300 series aircraft, dated Feb 8 2007 is a part of this STC. | 03.07.2007 |
| 10016084 | ARC AVIONICS CORP. | EASA.IM.A.S.02269 | | FAA ST03356AT | BOEING 737-200 BOEING 737-300 | Installation of a dual Universal Avionics UNS-1F FMS sys on Boeing 737-300 aircraft Installation includes: 2 Nav Control Units (NCU) in the EE bay and 2 GPS antennas. Center pedestal pilot and co-pilot Control Display Units (CDU). Nav data can be select (...) | | ARC Avionics Master Data List 750-120605, Revision G, dated November 10, 2006, or later approved revision. | The aircraft have to be equipped with a Universal Avionics EFIS as it provides the Navigation Displays and Data Concentrator Units M integrate the aircraft sensors and communications. FAA Approved Flight Manual Supplement, ARC Avionics document 730- (...) | 31.05.2007 |
| 10016085 | ARC AVIONICS CORP. | EASA.IM.A.S.02270 | REV. 2 | FAA ST03362AT | BOEING 737-200 BOEING 737-300 | Installation of a dual Universal Avionics UNS-1F Flight Management System (FMS) or UNS-1F WAAS capable FMS, on Boeing 737-200 and 737-300 aircraft without EFIS (Configuration I) - Boeing 737-300 aircraft with original Honeywell or Collins EFIS (Config (...)) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. - CRI H-01 issue 3 dated March 07, 2011: Special condition - Enhanced Airworthiness Progr (...) | Master Data List: MDL 752-120605 Rev C, dated April 02, 2009 (B737-200 and B737-300 A/C without EFIS) - MDL 752-022310 Rev B, dated June 30, 2010 (B737-300 A/C with EFIS) / FAA approved Flight Manual Supplements: AFMS 732-022310, Rev B.2 dated 2011 (B737 (...)) | none | 28.04.2011 |
| 10016086 | ARC AVIONICS CORP. | EASA.IM.A.S.02271 | | FAA STC ST03363AT | BOEING 737-200 BOEING 737-300 | Installation of a single Universal Avionics Class #A# TAWS system on 737-200 series aircraft. | | - ARC Avionics Master Data List 762-061505, Rev #I/R, dated November 1, 2006, or later approved revision. - Flight Manual Supplement Doc 732-061505 Rev A, dated Feb 08 2007, or later approved revision. | The aircraft has to be equipped with a Universal EFIS to display in the center pedestal in order to display the TAWS terrain. This MFD can be part of this installation, or pre-existing as the 737-200 series Aircraft. EASA Approved Flight manual Supp (...) | 27.06.2007 |
| 10016298 | ARC AVIONICS CORP. | EASA.IM.A.S.02595 | | FAA STC ST03458AT | BOEING 737-300 BOEING 737-400 BOEING 737-500 | Installation upgrade to replace the existing system with either Honeywell or Collins Dual Enhanced Mode S Transponder system. The existing Omni directional antenna is compatible with the new system and will not be replaced. The two Honeywell TRA-67A (...) | | - ARC Avionics Master Data List 750-030806, rev H, dated march 30, 2007 or later EASA approved revision in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this Decision). - AFM Document, ARC Avionics Document 730-030806, for B (...) | - AFM Document, ARC Avionics Document 730-030806, for Boeing 737-300, -400, -500 series aircraft, dated 2 July 2007, or later EASA approved revision. | 10.01.2008 |
| 10034376 | ARCHEION HOLDINGS L.L.C. | | | FAA STC ST03529AT | 1. B757-200 2. B757-300 | Installation of an Avionics satLINK Communication Management System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. For changed areas the certification basis is CS 25 amendment 2<(-, <-)>25.301, 25.303, 25.305, 25.3 (...) | Archeion Holdings LLC Master Drawing List doc. N° 27005-MDL revision G dated 18th Aug. 2010 AFM Supplement, 072950-503, Revision C, 28th march 2011 ICA 27005-ICA, revision F, 25th June 2010 or later revisions of the above listed documents approved by EAS (...) | None | 30.03.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------|-------------------|----------------|----------------------|---|--|---------------------|--|--|---------------|
| 10015815 | ARINC INC. | EASA.IM.A.S.01886 | | FAA STC SA00558DE-D | BEECH 200, 200C, B200, B200C, BEECH 300, 300LW, B300, B300C | Installation of a Pilot and Co-Pilot Digital Air Data Unit | | Master Data List, document No 34-00-0002-00, Revision IR, dated 23 July 2004 or later approved revision/Instructions for Continued Airworthiness, document 34-02-0002-00, Rev 4.0 dated 16 July 2004 or 34-02-0002-10 Rev 1.0 dated 28 June 2005 or later approved revision. | 1. Airplane Flight Manual Supplement: AFMS Document 34-07-0002-00, Revision IR, dated 19 August 2004 or Document 34-07-0026-00, Revision IR or later approved revision (required onboard for the ISAS ADDU installation, or AFMS Document 34-07-0002-01, R (...)) | 02.01.2006 |
| 10015823 | ARMSTRONG AEROSPACE | EASA.IM.A.S.01895 | | FAA STC ST01872CH | B737-200, 737-300, 737-400, B737-500, 737-600, 737-700, B737-800, 737-900 | Installation of a Honeywell RESCU 406AF Emergency Locator Transmitter (ELT) System on Boeing 737 Series aircraft | | - Armstrong Aerospace Master Data List No DC104-1005-00, Revision G, dated 16 February, 2006, or later approved revision.- EASA approved AFM Supplement, Armstrong Aerospace Doc No DC303-1005-00, Revision C, (FAA approved April 15, 2005), or lat (...) | Compatibility of this design change with other previously approved modifications must be determined by the installer. | 25.04.2006 |
| 10015590 | ARMSTRONG AEROSPACE Inc. | EASA.IM.A.S.01501 | | FAA STC ST01742CH | BOEING 747-400 SERIES BOEING 747-400D BOEING 747-400F | Installation of Honeywell RESCU 406AF Emergency Locator Transmitter (ELT) system | | Armstrong Aerospace master Data List Document DC104-1011-00 Revision M dated March 25, 2005 or later approved revision/Armstrong Aerospace AFM Supplement Document number DC303-1011-00 Revision C approved April 20, 2005 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.10.2005 |
| 10015662 | ARMSTRONG AEROSPACE Inc. | EASA.IM.A.S.01639 | | FAA STC ST01773CH | BOEING 777-200 BOEING 777-300 | Installation of Honeywell RESCU 406AF Emergency Locator Transmitter (ELT) | | Armstrong Aerospace Master Data List Doc No DC104-1001-00 Revision A dated February 17, 2005 or later approved revision/Armstrong Aerospace AFM Supplement DC303-00 Revision - approved February 18, 2005 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.10.2005 |
| 10015664 | ARMSTRONG AEROSPACE Inc. | EASA.IM.A.S.01640 | | FAA STC ST02108CH | BOEING 767-300 | Installation of Honeywell RESCU 406AF Emergency Locator Transmitter (ELT) | | Armstrong Aerospace Master Data List Doc. No DC104-3058-00 Revision A dated January 19, 2005 or later approved revision. Armstrong Aerospace AFM Supplement Doc. No DC303-3058-00 Revision approved February 3, 2005 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.11.2005 |
| 10015725 | ARMSTRONG AEROSPACE Inc. | EASA.IM.A.S.01737 | | ST02202CH | BOEING 757-200 | Installation of Honeywell RESCU 406AF Emergency Locator Transmitter (ELT) | | Armstrong Aerospace Master Data List Doc No. DC104-3078-00-Revision B dated November 09 2005 or later approved revision. Armstrong Aerospace AFM Supplement Doc No. DC303-3078-00 Revision - approved November 04 2005 or later approved revision. | | 26.01.2006 |
| 10015789 | ARMSTRONG AEROSPACE Inc. | EASA.IM.A.S.01845 | | ST01797CH | AIRBUS A320 AIRBUS A321 | Installation of a Honeywell RESCU 406 AF Emergency Locator Transmitter (ELT) | | Master Data List (MDL), ref. DC104-1020-00, revision F, dated June 17, 2005 or later EASA approved revision. Airplane Flight Manual Supplement (AFM-S), ref. DC303-1020-00, revision A, dated February 4, 2004 or later EASA approved revision. | Instructions for Continued Airworthiness (ICA), ref. DC113-1020-00, Section 2, revision D, dated June 17, 2005 or later EASA approved revision | 17.05.2006 |
| 10016211 | ARMSTRONG AEROSPACE Inc. | EASA.IM.A.S.02458 | | FAA ST02404CH | AIRBUS A318 AIRBUS A319 AIRBUS A320 AIRBUS A321 | Activation of Honeywell Runway Awareness and Advisory System (RAAS). | | FAA STC ST02404CH, issued 4th April 2007/Armstrong Aerospace Master Data List, ref. Document No. DC104-4042-00, dated 21.03.07 or later revisions approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision (...)) | | 31.10.2007 |
| 10016363 | ARMSTRONG AEROSPACE Inc. | EASA.IM.A.S.02685 | | FAA STC ST02501CH | AIRBUS B4-603 AIRBUS B4-608R | Activation of Honeywell Runway Awareness and Advisory System (RAAS)(Validation of the FAA STC n° ST02501CH) This STC consists in activating the RAAS function on EGPWS | | Master Data List, ref : DC104-5029-00 Revision B, dated 08 November 2007. (or any later approved revision of this document). Certification Review Item F-01, issue IR, dated 18 February 2008/Aircraft Flight Manual Supplement, ref : DC303-5029-00, dated (...) | This STC is only applicable on Honeywell EGPWS P/N identified through the 'Associated Technical Documentation' fitted with the Hardware/Software configuration up to ref : 226/226 | 13.05.2008 |
| 10016553 | ARMSTRONG AEROSPACE Inc. | EASA.IM.A.S.02935 | | FAA STC ST01837CH | BOEING 747-200B BOEING 747-200C BOEING 747-200F BOEING 747-300 | Installation of Honeywell RESCU 406AF/AFN Emergency Locator Transmitter (ELT) System. | | Armstrong Aerospace Master Data List, Doc. No. DC104-1003-00, Rev. R, dated 20 December 2007 or later approved revision. | Airplane Flight Manual Supplements Armstrong Aerospace Doc.No. DC303-1003-00, Rev.C, approved 4 February 2008 or later approved revision is required on board of the modified aircraft. Instruction for Continued Airworthiness, Armstrong Aerospace Doc. No (...) | 12.09.2008 |
| 10016554 | ARMSTRONG AEROSPACE Inc. | EASA.IM.A.S.02936 | | FAA STC ST02536CH | AIRBUS A330-301 AIRBUS A330-302 AIRBUS A330-303 AIRBUS A330-321 AIRBUS A330-322 AIRBUS A330-323 AIRBUS A330-341 AIRBUS A330-342 AIRBUS A330-343 | Activation of Honeywell Runway Awareness and Advisory System (RAAS). | | # FAA STC ST02536CH, issued 17th March 2008# Armstrong Aerospace Master Data List, ref. Doc. No. DC104-5030-00, Rev. A, dated 4th February 2008 or later revisions approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent rev (...)) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 03.09.2008 |
| 10016555 | ARMSTRONG AEROSPACE Inc. | EASA.IM.A.S.02937 | | FAA STC ST02537CH | AIRBUS A340-311 AIRBUS A340-312 AIRBUS A340-313 AIRBUS A340-642 | Activation of Honeywell Runway Awareness and Advisory System (RAAS) | | # FAA STC ST02537CH, issued 17th March 2008# Armstrong Aerospace Master Data List, ref. Doc. No. DC104-5031-00, Rev. A, dated 4th February 2008 or later revisions approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent rev (...)) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 08.01.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------|-------------------|----------------|----------------------|---------------------------------------|---|---|--|---|---------------|
| 10029877 | ARMSTRONG AEROSPACE Inc. | | | FAA STC ST02776CH | B737-300, B737-500 | Activation of Honeywell Runway Awareness and Advisory System (RAAS) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01/ Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane Systems" (...) | Armstrong Aerospace documents- Master Data List - DC104-7006-00 Rev D dated 10 Nov 2009-AFM Supplement document DC303-7006-00 approved on 04 Nov 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision (...) | None | 28.04.2010 |
| 10031966 | ARMSTRONG AEROSPACE Inc. | | | FAA STC ST02778CH | 747-400, 747-400 (BCF) 747-400F | Activation of Honeywell Runway Awareness Advisory System (RAAS) in accordance with Armstrong Aerospace Master Data List DOC No.DC104-7007-00 Revision D, dated November 9, 2009 or later FAA approved revision. Validation of FAA STC ST02778CH | The Certification Basis for the original product remains applicable to this certificate/ approval | or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 27.09.2010 |
| 10031995 | ARMSTRONG AEROSPACE Inc. | | | FAA STC ST02376CH | ATR72-201, ATR72-202, ATR72-212 | EASA Validation of FAA STC ST02376CH (as amended by FAA on 24th June 2010) regarding: Installation of a Rockwell Collins HF 9000 HF Radio System in accordance with Armstrong Aerospace Master Data List, Doc. Number DC104-4033-00, Revision E, dated 27th May 2010, or later approved revisions | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: An EASA Special Condition CRI H-001 Issue 01 "Enhanced Airworthiness Programme for Aeropl (...) | Armstrong Aerospace Master Data List, Doc. Number DC104-4033-00, Revision E, dated 27th May 2010, or later approved revisions. Armstrong Aerospace Doc. No. DC303-4033-00, Revision B (>,<,-) or later revisions of the above listed documents approved b (...) | None | 30.09.2010 |
| 10034747 | ARMSTRONG AEROSPACE Inc. | | | FAA STC ST02939CH-D | B737-800 | Validation of FAA STC ST02939CH-D PANASONIC DMPEs - In flight Entertainment System installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Cf. EASA TCDS IMA.A.120 and: CRI H-01 issue 01 on EWIS (>,<,-) CRI F-01 issue 01 on IFET (...) | Master Data List, Ref: DC104-8084-00 Rev B, dated 22 April 2011-Airplane Flight Manual Supplement, Ref: DC303-8084-00, dated 15 April 2011-Instruction for Continued Airworthiness Supplement, Ref: DC113-8084-00, Rev A, dated 15 April 2011 or later revisi (...) | None | 27.04.2011 |
| 10038436 | ARMSTRONG AEROSPACE Inc. | | REV. 2 | FAA STC ST02947CH | 737-800 | Validation of FAA STC ST02947CH installation and activation of a Panasonic Global Communications Suite (GCS) system in accordance with Armstrong Aerospace Master Data List DC104-9043-00, Revision M, dated 17 December 2012. Revision 1 of this STC has been (...) | CS-25 at amendment 2, except: -JAR 25.571 (CRI A11-05) as per TCDS (>,<,-) CRI H-01/EWIS at issue 3 dated 17 February 2012 (>,<,-) CRI WLAN at issue 3 dated 23 February 2012. - GSM CRI F-01 at issue 3 dated 03 January 2012 (>,<,-) Database CRI F-04 at (...) | Master Data List, Reference DC104-9043-00, Revision M, dated 17 December 2012. Instructions for Continued Airworthiness Supplement DC113-9043-00, Revision E, 23 July 2012. Weight and Balance supplement DC304-9043-00, Revision C, dated 16 May 2012. Airplane (...) | 1) Modification in accordance with this STC requires prior or concurrent modifications in accordance with Armstrong Aerospace Wireless and Cellular technologies "provisions" (EASA STC 10042528) and "Certification of compatibility with Cellular and Wlr (...) | 21.01.2013 |
| 10040305 | ARMSTRONG AEROSPACE Inc. | | | FAA STC ST03085CH | B747-400 | Installation of provisions for Panasonic Global Communications Suite (GCS) System in accordance with Armstrong Aerospace Master Data List DC104-9027-00 Rev. D dated April 25, 2012, or later approved revision. Validation of FAA STC ST03085CH | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Regulations at the latest amendment FAR Part 25-0 through 25-11925.23(a)-(b); 25.29(a) (...) | DC104-9027-00 Rev. D - Master Data List dated April 25, 2012 DC113-9027-00 - Instructions for Continued Airworthiness, section 2 Airworthiness Limitations, Rev. C, dated February 23, 2012 DC303-9027-00 Airplane Flight Manual Supplement or later revision (...) | None | 22.06.2012 |
| 10040470 | ARMSTRONG AEROSPACE Inc. | | | FAA STC ST02948CH | 767-300 | Installation of Board Connect In-Flight Entertainment System | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this approval. | Master Drawing List DC104-9024-00, Revision B, 22/01/2012 AFM Supplement DC303-9024-00, Revision -, 16/02/2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | This system is intended to provide internet connection and e-mails services using portable electronic devices (PEDs). Any other intended function of this equipment will require further certification approval. | 05.07.2012 |
| 10040513 | ARMSTRONG AEROSPACE Inc. | | REV. 1 | FAA STC ST03047CH | 747-400 | Certification of Compatibility with Cellular and Wireless Data Technologies. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at the later amendment: 25.1309(a) of CS-25 at amendment 2 and 25.1529 of CS-25 at amendment 2 T (...) | Armstrong Aerospace Master Data List DC104-9026-00, Revision C, dated 19 January 2013 Armstrong Aerospace Aircraft Equipment Compatibility List DC370-9026-01, Revision B, dated 18 January 2013 Armstrong Aerospace Aircraft Equipment Compatibility List DC (...) | The following are the limitations associated with this type design change. 1., This STC does not constitute installation approval of aeroplane hardware. It only provides data that may partially support their installations or activations. This STC may only (...) | 24.07.2013 |
| 10041316 | ARMSTRONG AEROSPACE Inc. | | REV. 1 | FAA STC ST03086CH | B747-400 | Installation and activation of a Panasonic Global Communication Suite (GCS) System (>,<,-) including the Panasonic eXPhone system in accordance with Armstrong Aerospace Master Data List Doc. No. DC 104-9034-00 Revision H, dated December 14, 2012. The add (...) | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: For the areas of change: the following requirements are complied with amendment 45 of FAA 14 CFR Part 25.571(a) - (b) For (...) | Certification Plan E11058-2500-C001, Rev. D (initial STC)- Certification Plan E12066-2500-C001, Rev. B- Armstrong Aerospace Master Data List: DC104-9034-00 Revision H, dated December 14, 2012 (>,<,-) Aircraft Maintenance Manual Supplement: Doc. No. (...) | Modification of the aircraft in accordance with this STC requires prior concurrent modifications in accordance with FAA STCs 10040513 (FAA STC ST03047CH) and 10040305 (FAA STC ST03085CH). FAA approved Airplane Flight Manual Supplement DC303-9034-00 (...) | 10.04.2013 |
| 10042075 | ARMSTRONG AEROSPACE Inc. | EASA.IM.A.S.00252 | REV. 2 | FAA STC ST01997CH | DC-9-83 (MD-83) | Installation of Honeywell Rescu 406AF/AFN system on DC-9-83(MD-83). This Rev. 2 is to correct a typo in the original STC number. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | DC104-3063-00 Rev. G dated 30 October 2012 Master Data List DC113-7000-00 dated 10 August 2009 Instruction for Continued Airworthiness: DC303-3063-00 Rev. C Aircraft Flight Manual Supplement: DC304-7000-00 dated 15 June 2009 Weight and Balance Manual S (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 26.03.2013 |
| 10042211 | ARMSTRONG AEROSPACE Inc. | | REV. 1 | FAA STC ST02377CH | A319-131, A319-133 A320-232, A321-232 | EASA validation of FAA STC ST02377CH "Installation of a Rockwell Collins Talkwind 560 Satellite System and modification of a Thales Top Series 3000/4000 In flight Entertainment (IFE) System" | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | See Armstrong Aerospace's documents- Master Data List Doc.No. DC104-4038-00, Revision C dated July 31st 2007 and- Airplane Flight Manual (AFM) Supplement Doc. No. DC303-4038-00 Revision A, FAA approved August 21st, 2007 and- Instructions for continue (...) | The FAA STC ST02377CH validated through EASA STC Approval 10042211 for Air Cargo Bay Equipment Rack Installation must be installed on aircraft prior or in conjunction with this STC design change. Prior to installation of this modification it must be d (...) | 20.08.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|----------------------|----------------|----------------------|---|---|---|---|---|---------------|
| 10042212 | ARMSTRONG AEROSPACE Inc. | | | FAA STC ST02314CH | A319-111,-112,-113,-114,-115 A319-131,-132,-133 A320-111,-211,-212,-214,-231 A320-232,-233 A321-231,-232 | EASA validation of FAA STC ST02314CH "Installation of an Aft Cargo Bay Equipment Rack" | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | See Armstrong Aerospace's documents:- Master Data List Doc. No. DC104-3095-00, Revision D dated April 09th 2007 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of (...)) | none | 15.11.2012 |
| 10042919 | ARMSTRONG AEROSPACE Inc. | | | FAA STC ST02946CH | 737-600, 737-700, 737-800 737-900 | Validation of FAA STC ST02946CH Certification of compatibility with Cellular and Wireless Data Technologies in accordance with Armstrong Aerospace DC104-9047-00, Revision F, dated 02 October 2012 | CS-25 at amendment 2, except:- WLAN CRI at issue 3 dated 23 February 2012,- GSM CRI F-01 at issue 3 dated 03 January 2012<->->The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | - Master Data List, Reference DC104-9047-00, Revision F, dated 02 October 2012- Aircraft Equipment Compatibility list, Reference DC370-9047-00, Revision F, dated 27 September 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of (...)) | 1) This STC may only be applied to an aircraft when an airplane installed base station using wireless and cellular technologies is concurrently installed in accordance with separate EASA approvals. 2) This STC does not constitute installation approval of (...) | 03.01.2013 |
| 10043422 | ARMSTRONG AEROSPACE Inc. | | | FAA STC ST03048CH | B777-200/200LR/300/300ER | Certification of Compatibility with Cellular and Wireless Data Technologies | The Certification Basis for the original product remains applicable to this certificate/ approval. | Armstrong Aerospace Master Data List DC104-9044-00, Revision E dtd. 21 December 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the (...) | 1) Honeywell Display Unit, P/N 4091900-xxx (Example P/Ns 4091900-931, 4091900-932, 4091900-933 and 4091900-934), has exhibited susceptibility during 4 watt EIRP Personal Electronic Devices (PEDs) wireless dataradio frequency compatibility tests 2) This (...) | 28.01.2013 |
| 10043586 | ARMSTRONG AEROSPACE Inc. | | | FAA STC ST013106CH | A330-201, A330-202, A330-203, A330-223, A330-243, A330-301, A330-302, A330-303, A330-321, A330-322, A330-323, A330-341, A330-342, A330-343. | Certification of Compatibility with Cellular and Wireless Data Technologies | The Certification Basis for the original product remains applicable to this certificate/ approval. | Armstrong Aerospace Master Data List DC104-9045-00, Revision A dated 20 December 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the (...) | 1) This STC does not constitute installation approval of aeroplane hardware. It only provides data that may partially support their installations or activations. This STC may only be applied to an aircraft when an airplane installed base station using (...) | 08.02.2013 |
| 10043936 | ARMSTRONG AEROSPACE Inc. | | | FAA STC ST03109CH | A330-201, A330-202, A330-203 A330-223, A330-243 A330-301, A330-302, A330-303 A330-321, A330-322, A330-323 A330-341, A330-342, A330-343 | Certification of Compatibility with Cellular and Wireless Data Technologies. | The Certification Basis for the original product remains applicable to this certificate/ approval. | Armstrong Aerospace Master Data List DC104-9209-00, Revision B dtd. 14 December 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the (...) | 1) This STC does not constitute installation approval of aeroplane hardware. It only provides data that may partially support their installations or activations. This STC may only be applied to an aircraft when an airplane installed base station using (...) | 06.03.2013 |
| 10016057 | AS&T AEROSPACE SYSTEM & TECHNOLOGIE | EASA, IM, A.S. 02234 | | FAA STC SA755GL | CESSNA 206H CESSNA T206H CESSNA TU206G CESSNA U206F CESSNA U206G | Installation of TKS Ice Protection Systems Components on a no-hazard / no credit basis. | | AS&T Drawing List No. 02800-DL, Revision 3, dated 19 December 2003, or later approved revision. #Supplemental Airplane Flight Manual and FAA approved Airplane Flight Manual Supplement, Doc. No. 02800-FMS, revised May 27, 2003 or later approved revision. | 1. This installation is not approved for flight into known icing conditions. 2. A placard must be fitted on the upper control panel in front of the pilot, stating #FLIGHT INTO KNOWN ICING CONDITIONS IS PROHIBITED#. 3. This installation is approved on (...) | 04.01.2007 |
| 10016058 | AS&T AEROSPACE SYSTEM & TECHNOLOGIE | EASA, IM, A.S. 02235 | | FAA STC SA01338WI | CESSNA 182P CESSNA 182Q CESSNA 182R CESSNA 182S CESSNA 182T CESSNA R182 CESSNA T182 CESSNA TR182 | Installation of TKS Ice Protection Systems Components on a no-hazard / no credit basis | | AS&T Master Drawing List No. 12500-DL, Revision 5, dated May 27, 2005 or later approved revision. #Supplemental Airplane Flight Manual and FAA approved Airplane Flight Manual Supplement, Doc. No. 12500-FMS, dated 27-Sep-2005 or later approved revision. | 1. This installation is not approved for flight into known icing conditions. 2. A placard must be fitted on the upper control panel in front of the pilot, stating #FLIGHT INTO KNOWN ICING CONDITIONS IS PROHIBITED#. 3. Applicable serial numbers: 182P, S/ (...) | 08.12.2006 |
| 10016257 | AS&T AEROSPACE SYSTEM & TECHNOLOGIE | EASA, IM, A.S. 02535 | | FAA STC SA01328WI | PIPER PA-32-301FT (PIPER 6X) PIPER PA-32-301XTC (PIPER 6XT) PIPER PA-32R-301 PIPER PA-32R-301T | Installation of TKS Ice Protection Systems Components on no-hazard / no credit basis. | | Installation must be in accordance with AS&T Master Drawing List No. 12700DL, Revision 3, dated March 7, 2005 or later approved revision. Operation must be in accordance with Supplemental Airplane Flight Manual and FAA approved Airplane Flight Manual (...) | 1. This installation is not approved for flight into known icing conditions. 2. A placard must be fitted on the upper control panel in front of the pilot, stating #FLIGHT INTO KNOWN ICING CONDITIONS IS PROHIBITED#. 3. This installation is approved on a no-hazard / no credit basis. | 27.08.2007 |
| 10031933 | ASI INNOVATION | | | | T206H | ASI-FAM-0009-01 Installation of FLJR MX10 mechanical provision on T206H aircraft | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/ approval. | None | Limitation to Cessna T206H serial number 445 and 446 | 23.09.2010 |
| 10032904 | ASI INNOVATION | | | | F406 | ASI-FAM-0015-02 Provision installation of console PMR/DMR and UHF antenna | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | None | Limited to: F406 S/N 006-017-025-039-042-070-074-075-077-086 | 07.12.2010 |
| 10035229 | ASI INNOVATION | | | | T206H | ASI-FAM-0002-01 Installation of Flir MX10 on Cessna T206H. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Aircraft Flight Manual Supplement ASI-AFM-0002-03 dated 22 April 2011 ASI-SDL-0002-05 Maintenance Planning Supplement ASI-MPS-0002-01 dated 17 March 2011 or later revisions of the above listed documents approved by EASA. | Limitation to Cessna T206H serial numbers 445 and 446. | 07.06.2011 |
| 10039405 | ASI INNOVATION | | | | CESSNA F150L, FA150L | ASI-FAM-082-04 - Installation of EFIS Aspen EFD 1000 on Cessna 150 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | None | The EFD 1000 must not be used as primary flight and navigation source. Independent and dissimilar instruments must be installed for primary indication of attitude, altitude, airspeed, vertical speed, magnetic heading and navigation data, as required for day and night VFR flights. | 02.05.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------|-------------------|----------------|----------------------|--|---|--|---|---|---------------|
| 10040616 | ASI INNOVATION | | | | 1. C90, C90A, C90GT, C90GTI 2. B90, E90 | VHF, UHF and GPS antennas installation provisions according to ASI modification no ASI-FAM-0096-00 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement no ASI-AFM-0096, original issue dated 20 June 2012. STC Data List no ASI-SDL-0096, revision 0 dated 25 June 2012 or later revisions of the above listed documents approved by EASA. | This change is applicable to aircraft: B90 S/N LJ-318 to LJ-501-C90 S/N LJ-502 to LJ-1062-C90A S/N LJ-1062 to LJ-1726, LJ-1727 to LJ-1753 and LJ-1754-C90GT S/N LJ-1727, LJ-1754, LJ-1756 and after-C90GTI S/N LJ-1847, LJ-1853 and after-E90 S/N LW-1 and after equipped with standard 250 VA Inverter | 13.07.2012 |
| 10040891 | ASI INNOVATION | | | | P2006T | ASI-0084-03 dated 30 July 2012 - Installation of removable mission console (without equipments) in the aircraft cabin in place of passenger seat. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/ approval. | Airplane Flight Manual Supplement ASI-AFM-0084-00 dated 07 June 2012. STC Data List ASI-SDL-0084-03 or later revisions of the above listed documents approved by EASA. | This STC shall be used only by ASI Innovation. | 01.08.2012 |
| 10040999 | ASI INNOVATION | | | | EMB-135LR | Oven installation. Description as per: ASI SDL STC Data List ref. ASI-SDL-0087-03, dated 01/08/2012. ASI MDL, Master Drawing List ref. ASI-MDL-0087-01, dated 01/08/2012. ASI Technical Dossier ref. ASI-TD-0087-02, dated 01/08/2012. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ASI SDL, STC Data List ref. ASI-SDL-0087-03, dated 01/08/2012. ASI MDL, Master Drawing List ref. ASI-MDL-0087-01, dated 01/08/2012. ASI Technical Dossier ref. ASI-TD-0087-02, dated 01/08/2012 or later revisions of the above listed documents approved by EASA. | Applicability limited to EMB-135LR S/N 886 | 14.08.2012 |
| 10041603 | ASI INNOVATION | | | | EMB-120ER | "P-RNAV Certification" Description as per documents in "Associated Technical Documentation". | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | STC data list reference ASI-SDL-0089-08, dated 26/09/2012. Technical Dossier ASI-TD-0089, issue 4, dated 26/09/2012. AFM Supplement ASI-AFM-0089, issue 6, dated 18/09/2012 or later revisions of the above listed documents approved by EASA. | Applicability restricted to the following S/Ns: 012, 013, 027, 158. | 01.10.2012 |
| 10041778 | ASI INNOVATION | | | | EMB-135LR | Cabin interior modification. Description as per Associated Technical Documentation. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ASI SDL, STC Data List ref. ASI-SDL-0059-01, dated 12/10/2012. ASI MDL, Master Drawing List ref. ASI-MDL-0059-02, dated 01/10/2012. ASI Technical Dossier ref. ASI-TD-0059-01, dated 09/10/2012. ASI AFMS ref. ASI-AFM-0059 issue 0, dated 27/06/2012 or lat (...) | Applicability limited to EMB-135LR S/N 886. | 15.10.2012 |
| 10042045 | ASI INNOVATION | | REV. 2 | | E90 | Mast with relay antenna installation. Revision : Update of drawings on ASI-MDL-0101-07 (-> -3)-Extension of applicability to all MSN on ASI-AFM-0101-01 and Update of above technical documentation on SDL ASI-SDL-0101-03 to reflect the changes. Revision 2 (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ASI-FAM-0101-03 dated 23/07/2013. ASI-SDL-0101-03 dated 23/07/2013. ASI-AFM-0101-01 dated 24/06/2013 or later revisions of the above listed documents approved by EASA. | The approval holder shall fulfill the obligations of Part 21, Paragraph 21A.109. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or (...) | 28.08.2013 |
| 10042513 | ASI INNOVATION | | REV. 4 | | ATR 42-300, ATR 42-500 ATR 72-201 | Radio Honeywell HF 950 Installation. This modification consists in installing the Bendix/King KHFS950 HF/SSB Communication System. This includes mainly the embodiment of HF antenna, control unit, receiver, amplifier coupler and electrical protection. Rev. (...) | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: ATR Special Condition H-01: Instructions for Continued Airworthiness for Electrical Wiring Interconnection System (EWIS). C (...) | STC Data List (ATR 42-300), ref : ASI-SDL-0095 Issue 04, dated 04 December 2012. Aircraft Flight Manual Supplement (ATR 42-300), ref : ASI-AFM-0095, dated July 2012. Instruction for Continued Airworthiness (ATR 42-300), ref : ASI-MPS-0095 Revision 01, dated (...) | This STC is limited to : MSN 148, MSN 227, MSN 559, MSN 643 and MSN 667. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair (...) | 07.06.2013 |
| 10043741 | ASI INNOVATION | | | | B300 | Replacement of the Universal FMS UNS-1M by the FMS UNS-1EW including P-RNAV capability. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Service Bulletin, ASI Innovation document ASI-SB-B300-3400-0106, Revision 01, dated February 13, 2013. Aircraft Flight Manual Supplement, ASI Innovation document ASI-AFM-0106, Revision 0, dated October 2012. Instructions (...) | Limited to Serial Number FL-120. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon (...) | 21.02.2013 |
| 10045599 | ASI INNOVATION | | | | L-1329 (JETSTAR 731) | Replacement of Flight Management System in accordance with Service Bulletin ASI-SB-L1329-3400-0135 Rev. 2 | 14 CFR Part 25, Amdt 136, IAW ASI-CCL-0135-01 | Service Bulletin ASI-SB-L1329-3400-0135 Rev. 2. AFM Supplement ASI-AFM-0135-00. CAs ASI-MPS-0135-00. Technical File ASI-SDL-0135-02 or later revisions of the above listed documents approved by EASA. | Approval Limited to Airplane LOCKHEED L-1329 MSN 5092. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce n (...) | 05.07.2013 |
| 10046050 | ASI INNOVATION | | | | F406 | ASI-FAM-0010-01 - Replacement of the existing BNBL turret by the MX10 turret. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Supplément Manuel de Maintenance ASI-MPS-0010-03 du 25 juillet 2013. Supplément Manuel de Vol ASI-AFM-0010-00 du 24 novembre 2010 or later revisions of the above listed documents approved by EASA. | Limitation to F406 serial number 0090. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 08.08.2013 |
| 10016994 | ASI SERVICES | EASA.IM.R.S.01403 | | FAA STC SH296NM | BELL 206L-1 | Installation of a Rolls Royce model 250-C30P engine and associated airframe and tail rotor control modifications similar to the Model 206L-3. This STC has been classified as #BASIC# by the FAA and has been approved under the PTVP. | | Air Services International FAA Sealed Master Drawing List: ASI-206L-30P revision 8 dated 20 June 2002 or later approved revision # Air Services International Installation Manual: ASI-206L-1-FM revision 15 dated 3 January 2006 or later approved revision # Rotorcraft Flight Manual | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 06.06.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|---------------------|----------------|----------------------|------------------------------|---|---|---|---|---------------|
| 10046011 | ASPEN AVIONICS, Inc. | | REV. 1 | FAA STC SA 10822SC | SEE EASA APPROVED MODEL LIST | Installation of the EFD1000/EFD500 Evolution Flight Display System. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Installation in accordance with Aspen Avionics Installation Manual document 900-00003-001, Revision AQ.-Operation in accordance with Aspen Avionics Airplane Flight Manual Supplement document 900-00008-001, Revision S, dated August 2013.- Instructi (...) | Aircraft registered in countries under EASA authority are not eligible for installation of the CG100 "Connected Gateway" unit under FAA STCSA 10822SC at this time. Prior to installation of this design change it must be determined that the interrelationship (...) | 19.08.2013 |
| 10015758 | ASSOCIATED AIR CENTER | EASA.IM.A.S.01793 | | FAA STC ST10546SC | AIRBUS A319-115 | VIP Cabin modification | | Master Document List (MDL), ref. 2350000000, revision A, dated June 8, 2006 or later EASA approved revision. Airplane Flight Manual Supplement (AFM-S), ref. 2350021090, revision IR, dated June 9, 2006 or later EASA approved revision. Instructions for (...) | This STC is limited to the A319 CJ MSN 2592. Instructions for Continued Airworthiness (ICAW), Chapter 3 "Airworthiness Limitations" ref. 2350050090, revision A dated June 8, 2006 or later EASA approved revision. | 13.06.2006 |
| 10015490 | ASSOCIATED AIR CENTER, L.P. | EASA.IM.A.S.01252 | | FAA STC ST10474SC | AIRBUS A319-133 | Installation of an Executive Style Interior acc.: MDL 2300000091 Rev. D | | AFM Supplement: 2300021090 Rev. IR or later EASA approved/accepted revisions. AMM Supplement/Airworthiness Limitations/ICA 2300050090 | 1) Ref.: 2300021090 Rev. IR2) Limited to S/N 2192. Note: The airplane is limited to private use only unless the cabin door is rendered inoperable. | 21.11.2005 |
| 10015924 | ASSOCIATED AIR CENTER, L.P. | EASA.IM.A.S.02059 | REV. 2 | FAA STC ST10593SC | AIRBUS A319-115 | Revision to EASA STC reference EASA.IM.A.S.02059 - Introduction of revised Instructions for Continued Airworthiness. STCEASA.IM.A.S.02059 - Installation of an Executive Style Interior (reference FAA STC ST10593SC dated 6 February 2007). First installa (...) | | - Master Document List, Document Number 2388000000, Revision A, dated 5 February 2007, or later EASA approved revision.- Instructions for Continued Airworthiness, including Airworthiness Limitations Section, Document Number 23880050090, Revision B, d (...) | 1. The aircraft is limited to Private use only (not to be operated for hire, reward or common carriage). 2. This STC is approved only for the product configuration as defined in the above Master Document List. 3. Limitations and Conditions specified i (...) | 14.05.2009 |
| 10015975 | ASSOCIATED AIR CENTER, L.P. | EASA.IM.A.S.02126 | | FAA STC ST10613SC | AIRBUS A319-115 | Executive style interior (validation of FAA STC ST10613SC) | | Index List ref.: 2393000001, dated March 8, 2007 or later approved revision. Master Drawing List, ref.: 2393000030E, dated March 13, 2007 or later approved revision. Airplane Flight Manual Supplement, ref.: 2393021090E, dated March 13, 2007 or later approved revision | Instruction for Continued Airworthiness, Chapter 3 "Airworthiness Limitation", ref. 2393050090, dated March 8, 2007 or later approved revision | 19.03.2007 |
| 10016150 | ASSOCIATED AIR CENTER, L.P. | EASA.IM.A.S.02352 | | FAA STC ST10536SC | AIRBUS A319-115 | Installation of an Executive Style Interior in accordance with Master Document List, Document Number 2459000000, Revision IR, dated 25 October 2007, or later EASA approved revision. First installation is on MSN 2837 | | AFM Supplement, Document Number 2459021090, Revision IR, dated 25 October 2007, or later EASA approved revision. Instructions for Continued Airworthiness, including Airworthiness Limitations Section, Document Number 2459050090, Revision B, dated 24 October 2007, or later EASA approved revision. | The aircraft is limited to Private use only (not to be operated for hire, reward or common carriage). This STC is approved only for the product configuration as defined in the above Master Document List. Limitations and Conditions specified in FAA STC ST10536SC D remain applicable | 30.10.2007 |
| 10027816 | ASSOCIATED AIR CENTER, L.P. | P-EASA.IM.A.S.03147 | REV. 1 | FAA STC ST10895SC-D | A319-115 | Major Change to the EASA STC 10027816 (FAA STC ST10895SC-D) to introduce the Decrane's side facing divan (FAA STC ST101450CH) within the "Executive Style Interior" configuration. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.- CRI D-4301-004/Deviation "Installation of Interior Doors"- CRI D-4302-004/Special Co (...) | -, FAA Supplemental Type Certificate ST10895SC-D amended 10th May 2012.-, Master Document List, ref. doc. 2578000000, Revision C, dated 10th May 2012.-, Airplane Flight Manual Supplement, ref. doc. 2578021090, Revision IR, dated 21st September 2009.- I (...) | This approval is limited only to the installation made in Airbus Model Number A319-115, Serial Number 3133 | 11.07.2012 |
| 10030326 | ASSOCIATED AIR CENTER, L.P. | | | FAA STC ST10900SC-D | A319-115 | Installation of an Executive Interior | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.- CRI A-4301-007 - EASA Supplemental Type Certification Basis for AAC Installation of Ex (...) | - FAA STC ST10900SC-D, dated 8th June 2010- Master Document List, ref. doc. No. 2731000000, Revision IR, dated 8th June 2010- Airplane Flight Manual Supplement, ref. doc. No. 2731021090, Revision IR, dated 8th June 2010- Instruction for Continued Airwo (...) | This approval is limited to only the installation made in Airbus Model Number A319-115, Serial Number 3957. Airworthiness Limitation inspection to be performed in all the locations of the Cabin Interior Primary Structure Elements attachments for every 300 (...) | 10.06.2010 |
| 10035731 | ASSOCIATED AIR CENTER, L.P. | | REV. 1 | FAA STC ST10906SC-D | A319-115 | Installation of an Executive Style Interior EASA validation of FAA STC ST10906SC-D | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. EASA CRI A-4301-009 (STC Certification Basis) The requirements for environmental protect (...) | Master Data List doc. N° 2634000000 Rev. B dated July 19, 2011. Instructions for Continued Airworthiness doc. N° 2634050090 Rev. A dated December 22, 2010. Airplane Flight Manual Supplement doc. N° 2634021090 Rev. B dated July 18, 2011. EASA Compliance C (...) | restricted to MSN 3542 | 19.07.2011 |
| 10035915 | ASSOCIATED AIR CENTER, L.P. | | | FAA STC ST10903SC-D | A319-133 | Modification of an Executive Interior EASA Validation of FAA STC ST10903SC-D | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. EASA CRI A-4301-008 (STC Certification Basis) The requirements for environmental protec (...) | Master Data List doc. N° 2708000000 Rev. IR dated October 27, 2010. Instructions for Continued Airworthiness doc. N° 2708050090 Rev. IR dated September 10, 2010. Airplane Flight Manual Supplement doc. N° 2708021090 Rev. IR dated October 27, 2010. Enginee (...) | Restricted to MSN 910 | 27.07.2011 |
| 10036081 | ASSOCIATED AIR CENTER, L.P. | | REV. 2 | FAA STC ST10897SC-D | B737-700 | Change to EASA STC 0010036081 (Installation of an Executive Interior on B737700GW aircraft : Project No AAC-08-06-DAS). Installation of the ERDA Model 4402 Single occupant Side facing divan. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI A-01 Issue 3 dated April 18, 2011: EASA Supplemental Type Certification Basis for As (...) | - Certification Plan, Reference 2964000090, initial revision, dated 27 October 2011- Airplane Flight Manual Supplement, Reference 2582021090, Revision A, dated 12 November 2010- Master Data List, Reference 2582000000, Revision C, dated 19 April 2012- I (...) | 1) This approval is limited to Aircraft B737-700, Serial number 36027. 2) The aircraft operator must perform the instructions for continued airworthiness and comply with the Airworthiness Limitation Items as per the Instructions Continued Airworthiness, (...) | 29.06.2012 |
| 10042309 | ASSOCIATED AIR CENTER, L.P. | | REV. 3 | FAA STC ST10913SC | A319-133 | Validation of FAA STC ST10913SC - Installation of an Executive interior-Revision of the AFMS-Update of the MDL | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI D-4302-010 - "Isolated compartments" The requirements for environmental protection and the associated certified noise (...) | -FAA Supplemental Type Certificate ST10913SC, issued November 16, 2012.- Master Document List, reference document 2925000001 Rev. D, dated January 31, 2013.- JAR Part 25 Compliance Check List, reference document 2925000099 Rev. IR, dated November 21, 2012 (...) | This modification is applicable to MSN 4992. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse (...) | 01.03.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|----------------------|---|---|--|--|---|---------------|
| 10016730 | ASTEC HELICOPTER SERVICES AS | EASA.IM.R.S.01048 | | | S-61N | Installation of Honeywell MST 67A Mode S Transponder | | 1) Master Drawing List reference AHSE-51550 rev - or later approved revisions.2) Aircraft Flight Manual Supplement No AHS-042 dated 16.03.05 or later approved revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.11.2005 |
| 10017091 | ASTEC HELICOPTER SERVICES AS | EASA.R.S.01002 | REV. 1 | | AS 332L, AS 332L1, AS 332L2 | Skywatch SKY-497 Traffic Advisory System | | -Design Report AHSE-51151, Master Drawing AHSE-51150, Rev A, or later approved revision-Flight Manual Supplement AHS-014 | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.11.2005 |
| 10017092 | ASTEC HELICOPTER SERVICES AS | EASA.R.S.01003 | REV. 1 | | | Techtest CPI-503 ELT | | Design Report AHSE-51501, Master Drawing List AHSE-51500 Revision F, Flight Manual Supplement AHS-034Service Bulletin AHSB-2566-031, latest approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 15.11.2005 |
| 10038238 | ASTRONICS - LSI | | REV. 1 | FAA STC SA02387AK | 1- 58, 58A, G58, 95-B55, E55 2- A36, G36, A36TC, B36TC, 2- F33A, F33C | Fabrication and Installation of Exterior LED Lighting Suite in accordance with FAA STC SA02387AK. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: as referenced in FAA STC SA02387AK. The requirements for environmental protection and the associated certified noise and/or (...) | Fabrication and Installation of the LSI Exterior LED Lighting Suite in accordance with Astronics Master Data List (MDL) Doc. No. 23 16 00, Revision R7, dated November 5, 2012; Installation Instructions, One Sky Aviation, LLC document number 32 1605, Rev (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 30.01.2013 |
| 10016500 | ASTRONICS ADVANCED ELECTRONIC | EASA.IM.A.S.02862 | | FAA STC ST02535AT | BOEING 747-400 | Installation of a General Dynamics EmpPower AC In-Seat power Supply System on a Boeing 747-400 type aircraft in accordance with Master Drawing list MDL1091-3, Rev. C, dated August 6, 2002, or later FAA approved revision | | FAA approved Airplane Flight Manual Supplement Document Number D1091-517-8, dated June 6, 2002 or later approved revision | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 23.07.2008 |
| 10014147 | ATELIER CHABORD | EASA.A.S.01266 | REV. 1 | | DR 300/108, DR 315, DR 400/100, DR 400/120, DR 400/120 D, DR 400/2+2. | STCDR420-2009-01 Rev H - Extension applicabilité STC EASA.A.S.01266 et grille de protection de réchauffage carburateur | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 23 Amdt 2 This certificate/ approval involves a change to the requirements for (...) | Supplément manuel de vol Supmdv-dr420-09-01B dated 29 nov 2012 or later revisions of the above listed documents approved by EASA. | STC applicability shall be checked with installation and maintenance manual dated 03 jan 2013 and supplement flight manual. Prior to installation of this design change it must be determined that the interrelationship between this design change and any (...) | 25.01.2013 |
| 10037356 | ATELIER CHABORD | | | | F150F, F150G, F150H, F150J F150K, F150L, F150M FA150K, FA150L, FA150M | STC CESSNA150-11-01 Rev E - Dispositif de boîte de réchauffage carburateur additionnel sur ensemble collecteur d'échappement et silencieux. | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the certificated noise and/or emissions levels. | FAM STCCESNA150-11-01 rev E du 21 novembre 2011 Programme d'entretien additionnel entretien-cessna150-11-01 rev C du 31 oct 2011 or later revisions of the above listed documents approved by EASA. | Tout Reims Aviation Cessna F150F a M, FA150K a M modifié selon le STC38-SF-0001 ou toute révision ultérieure. | 22.11.2011 |
| 10039166 | ATELIER CHABORD | | REV. 1 | | CESSNA 172R, 172S | Exhaust system and silencer for general aviation aeroplanes | CS 23-(>,-<)> Amdt: 2 The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Instructions de montage et programme d'entretien Ref. montage-Cessna 172-1A (-2A, -3A, -4A) or later revisions of the above listed documents approved by EASA | None | 23.04.2012 |
| 10043168 | ATELIER CHABORD | | | | 152, F152, FA152 | STCCESNA152-12-01 rev C dated 08 Jan 2013 - Exhaust system and silencer installation. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 23CS23 Amdt 2 This certificate/ approval involves a change to the certificated noise and/or emissions levels. | Supplément au manuel de vol SUPMDV-CESNA152-12-01 Rev A dated 7 Dec 2012 or later revisions of the above listed documents approved by EASA | This STC is approved for aeroplanes equipped with Lycoming O-235-L2C or n2C and McCauley 1A103/TCM6958 propeller or Sensenich 72CK56-0-56 or -54 propeller. | 09.01.2013 |
| 10045818 | ATI ENGINEERING SERVICES, LLC | | | FAA SR02443NY-D | S-92A | Night Vision Goggle (NVG) compatible Lighting System | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS29 Issue 2: 29.1, 29.25(a), 29.610 (a) (b) (d), 29.771 (a), 29.773 (a) (2), 29.777, 29.853(a)(3)(4), 29.1301, 29.1309, (a) (...) | Rotorcraft Flight Manual FMS-100511 EASA rev. dated 24 June 2013 or later revisions of the above listed documents approved by EASA-INSTRUCTIONS FOR CONTINUED AIRWORTHINESS, document ICA-100511, Rev 2-INSTRUCTIONS FOR CONTINUED AIRWORTHINESS, EUROPE (...) | Installation of NVIS/NVG compatible lighting does not include or imply approval for flight operations with Night Vision Goggles (NVG). The operator must receive approval for the Civil Aviation operational authority to operate with NVG. The first install (...) | 19.07.2013 |
| 10016626 | ATLANTIC AERO INC | EASA.IM.A.S.03061 | | FAA STC ST00026AT | CESSNA 208 | Installation of Passenger Interior And Cargo Door Modification in Cessna Caravan (Cargo Conversion) | | Installation in accordance with Aero Modification and Consulting document 208199203 - Master Drawing List, Revision D, 2 August 2002, or later FAA approved revision. This STC is valid only with the FAA STC ST00026AT, and any EASA Form 96 Post-TC Technica (...) | - | 16.12.2008 |
| 10016627 | ATLANTIC AERO INC | EASA.IM.A.S.03062 | | FAA STC SA000154AT | CESSNA 208 CESSNA 208A CESSNA 208B | Installation of Cabin Windows in the Cargo versions of Cessna 208 series | | Installation will be done in accordance with Atlantic Aero document No. 208199204, Master Drawing List, Revision G, of 16 November 2006, or later FAA approved revision, and in accordance with the documents and limitations of FAA STC SA000154AT. | | 15.12.2008 |
| 10028652 | ATLANTIC AERO, Inc. | | | FAA STC ST02635AT | LEARJET 35, 35A, 36, 36A | Deactivation of Aerona (Learjet) Thrust Reversers | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Airplane Flight Manual Supplement 21193100, rev. I Instruction for Continued Airworthiness 21193010, rev. Cor later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | Not Applicable | 01.02.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|----------------|----------------|----------------------|---|--|---|--|---|---------------|
| 10023557 | ATLANTIC AERO, Inc. | | | FAA STC SA01539AT-D | 206, P206, P206A, P206B, P206D P206E, TP206A, TP206C, TP206D TU206C, TU206D, TU206E, TU206F TU206G U206, U206A, U206B, U206C, U206D, U206E, U206F, U206G, | Installation of a Remote Mounted Oil Cooler in accordance with AtlanticAero Inc. Master Drawing List 16693001, Rev. Basic, dated December 5, 1997 or later FAA approved revision in accordance with FAA STCSA01539AT-D | See FAA TCDS A4CEThe Certification Basis for the original product remains applicable to this certificate/ approvalThe certified noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval | -FAA approved Supplemental Airplane Flight Manual, Doc. No.16693102, Rev. Basic, dated February 4, 1999 for Cessna 206 (S/N 206-0001 through206-0275), U206 through U206D & TU206C through TU206D (S/N U206-0276through U206-1444), P206 through P206B and (...) | None | 01.04.2010 |
| 10041533 | ATLANTIC AERO, Inc. | | | FAA STC SA03931AT-D | TBM 700 A/B/C1/C2/N | Installation of modified battery heat shield | The Certification Basis for the original product remains applicable to this certificate/ approval.The certified noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | Atlantic Aero Instructions for Continued Airworthiness no 26493010revision IR dated 15 September 2011Atlantic Aero Master Drawing List no 26493001 revision IR dated 26September 2011for later revisions of the above listed documents approved by EASA | None | 25.09.2012 |
| 10014073 | ATLANTIC AEROENGINEERING LIMITED | EASA.A.S.01165 | | | BAE JETSTREAM 3100 | Installation of a Honeywell MK-VI GPWS & Honeywell RT 300 Radio Altimeter | | -Atlantic Aeroengineering Definition Document DD/AE/418 Issue 2 dated March 10 2005 or later approved revision-AAE Flight manual Supplementn AAE 6 Issue 1 or later approved revision | | 06.04.2005 |
| 10013952 | ATS INTERNATIONAL | EASA.A.S.00451 | REV 1 | | DC-9-80 (MD-80) SERIES | Mk VII EGPWS System Installation | | Master Data List, ref. 0197-01-A-2103 revision 08, dated January 25, 2006 or later EASA approved revision.Airplane Flight Manual Supplement, ref. 0197-01-A-2301 revision 03, datedMay 16, 2005 or later EASA approved revision.Airplane Flight Manual Su (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 21.03.2006 |
| 10013979 | ATS INTERNATIONAL | EASA.A.S.01004 | | | BOEING 737-200-200C | installation of a Class A Terrain Awareness Warning System (TAWS), based on an Honeywell EGPWS (S Mk VII) System, aboard Boeing 737-200aircraft equipped with external GPS source. | | MDL (Master Data List) : 0200-01-A-2103-R01. | AFMS : 0200-01-A-2301-R00.ADSS (Airworthiness Data Sheet Supplement) including ICA (Instructions for Continued Airworthiness): 0200-01-A-2203-R00.This STC is limited to Terrain display configuration 1 : Honeywell PPI-4B or Collins WX1 701/711. | 26.07.2005 |
| 10014033 | ATS INTERNATIONAL | EASA.A.S.01103 | | | BOEING 737-300 BOEING 737-400 BOEING 737-500 | Installation of an Artex Emergency Locator Transmitter | | - ATS International Master Data List No 0233-01-A-2103 revision 2- Airworthiness Limitations as specified in ATS- Airworthiness Data Sheet Supplement No 0233-01-A-2203 revision 02.6.5 | | 18.03.2005 |
| 10014053 | ATS INTERNATIONAL | EASA.A.S.01135 | | | 42-320-300 AEROSPATIALE ATR 42-300 AEROSPATIALE ATR 72-201 AEROSPATIALE ATR 72-202 AEROSPATIALE ATR 72-211 AEROSPATIALE ATR 72-212 ATR 72-101-102-201-202-211-212 | 121.5/243/406 Mhz Emergency Locator Transmitter Installation. | | ATS International Master Data List No. 0220-01-A-2103 Revision 1. | | 24.03.2005 |
| 10014120 | ATS INTERNATIONAL | EASA.A.S.01230 | REV 2 | | AIRBUS A318-319-320-321 | MK V EGPWS System installation | | Master Data List, ref. 0226-01-A-2103 revision 06, dated January 2, 2006 or later EASA approved revisionAirplane FlightManual Supplement (FMS source only), ref. 0226-01-A-2301 revision 03, dated July 21, 2005 or later EASA approved revisionAirplane (...) | | 01.03.2006 |
| 10014125 | ATS INTERNATIONAL | EASA.A.S.01235 | | | AIRBUS A319-320-321 | ELT antenna installation on A320 family | | Master Data list ref 0243-01-A-2103-R00, revision 0, dated March 24, 2003 | | 30.05.2005 |
| 10014182 | ATS INTERNATIONAL | EASA.A.S.01319 | | | BOEING 767-200 BOEING 767-300 | Installation of Enhanced Mode S Surveillance | | -Master Data List 0213-01-A-2103-R00-AFM Supplement 0213-01-A-2301-R01 | | 25.01.2006 |
| 10014183 | ATS INTERNATIONAL | EASA.A.S.01320 | | | B757 | Installation of emergency Locator Transmitter | | ATS International Modification STC 0248-01A2103-R00 Master Data List 0248-01A-2103-R00 | | 16.10.2006 |
| 10014235 | ATS INTERNATIONAL | EASA.A.S.01398 | | | BOEING 737-300 | ATLAS Standard Galleys Installation. | | ATS Master Data List, Doc. Ref. 4011-01-A-2103-R03, Revision 3 dated 31 August 2005 | | 18.10.2005 |
| 10014363 | ATS INTERNATIONAL | EASA.A.S.01631 | REV 1 | | DC-10 SERIES | Installation of a Honeywell Enhanced Ground Proximity Warning System MK VII. The FAA has processed a similar application for an STC for a similar installation under project n°: ST00536E. | | See item 5 of CAA Netherlands approval report STC ref. no. P-EASA.A.S.01631 dated 05/12/2006 | 1) Approval of this change in type design applies to the above model aircraft equipped with either a Flight Management System (FMS), which meets AC90-45A, a Global Positioning System (GPS) navigational based system or an EGPWS equipped with an inter (...) | 20.12.2006 |
| 10014402 | ATS INTERNATIONAL | EASA.A.S.01697 | | | BOEING 767-200 SERIES BOEING 767-300 SERIES | TCAS II Change 7 System Installation | | ATS International Master Data List 0267-01-A-2103-R00 or later approved revision.ATS International AFM Supplement 0267-01-A-2301-R03 or later approved revision. | | 22.05.2006 |
| 10014410 | ATS INTERNATIONAL | EASA.A.S.01706 | REV 1 | | DC-9-82 DC-9-83 (MD-83) | Elementary and Enhanced Surveillance Installation | | MDL 0269-01-A2103-R06AFMS 0269-01-A2301-R05 | | 27.02.2007 |
| 10014462 | ATS INTERNATIONAL | EASA.A.S.02522 | REV 1 | | | Mode S Elementary and Enhanced surveillance | | Master Data List (MDL) ref. 0264-01-A-2103 revision 05, dated June 19, 2006 of later EASA approved revisionAirplane Flight Manual Supplement (AFM-S) ref. 0264-01-A-2301 revision 04, dated June 19, 2006 of later EASA approved revision | | 30.06.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|----------------------|--|---|--|--|--|---------------|
| 10014503 | ATS INTERNATIONAL | EASA.A.S.02573 | | | AIRBUS A300 MODEL B2 AIRBUS A300 MODEL B4 | MK7 EGPWS Installation on A300 B2-B4Config 1: EGPWS with Dual PPI-910Config 2: EGPWS with Dual PPI-4BConfig 3: EGPWS with Dual PPI-4B and reactive Windshear | | - Config 1: Master Data List n° 0121-01-A-2103-R06Aircraft Flight Manual Supplement ref. 0121-01-A-2301-R05 or later EASA approved revision. Config 2: Master Data List n° 0121-02-A-2103-R02Aircraft Flight Manual Supplement ref. 0121-02-A-2301-R01 or (...) | Instructions for Continued Airworthiness are described in the Airworthiness Data Sheet Supplement ADSS 0121-01 - A-2203-R05 or later EASA approved revision. | 03.01.2006 |
| 10014762 | ATS INTERNATIONAL | EASA.A.S.02952 | | | A310 | Elementary and Enhanced Surveillance installation on A310 aircraft.This STC has been developed for the Mode S elementary/enhanced surveillance installation on A310 aircraft. | | System Technical Specification, ref : 0321-01-A-2003-R00Master Data List, ref : 0321-01-A-2103-R00(or any later approved revision)Aircraft Flight Manual Supplement, ref : 0321-01-A-2301-R00Instruction for Continued Airworthiness, ref : 0321-01-A-2211-R00 | This STC is approved only for the product configuration as defined in the approved design data referred to in paragraph#Description#.Compatibility with other aircraft/engine configurations shall be determined by the installer. | 02.04.2007 |
| 10042995 | ATS INTERNATIONAL | EASA.A.S.00502 | REV. 1 | | 737-300/400/500 | Installation of Honeywell EGPWS Mark VThis STC supersedes and replaces the DGAC STC No F-CIM135-SF-0708 | | -ATS International Master Data List No 0124-01-2103-R13-Configuration 1 (EGPWS with internal GPS); ATS International AFMSupplement No 0124-01-A-2301-R03-Configuration 2 (EGPWS with FMS); ATS International AFM Supplement No0124-01-R-2301-R02-Airworth (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description"Compatibility with other aircraft/engine configurations shall be determined by the installer. | 24.01.2005 |
| 10043013 | ATS INTERNATIONAL | EASA.A.S.01136 | | | 767-200,-300 | MK V EGPWS system installation. | | Alpine Flight Manual Supplement (AFMS) ref. 0203-01-A-2301 revision 0,dated June 13, 2005 or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The ins (...) | 23.10.2007 |
| 10014850 | AUSTRIAN AIRLINES | EASA.A.S.03072 | REV. 1 | | A320, A321 SERIES | Replace Airshow Digital Interface Unit (DIU) Model 210 by Model 420 This design change comprises the removal of Airshow210 Unit (P/N 910030 or P/N 910040) and installation of Airshow 420 Unit (P/N 921000, 921005 or 921006). | | Supplemental Type Definition Report, ref. STDR-A32F-23-0008-00, dated 02.08.2007 or later EASA approved revisions. Declaration of Compliance, ref. DC-A32F-23-0008-00, dated 02.08.2007 | | 21.01.2008 |
| 10015197 | AUSTRIAN AIRLINES | EASA.A.S.03587 | | | F28 MARK 0070, F28 MARK 0100 | Installation of mechanically timed water faucet in Fokker F28 Mark 0070, Fokker F28 Mark 0100. | | Supplemental Type Definition Report Nr: STDR-FKR-38-0001-02. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 31.03.2009 |
| 10015553 | AVCON DIVISION/BUTLER NATIONAL CORP | EASA.IM.A.S.01449 | | FAA STC SA2479CE | LEARJET 35, 35A, 36, 36A | Installation of dual openable hatches on the belly of the fuselage. | | Avcon Master Drawing List 1054 Revision A dated April 15 2005 or later approved revision.AFM Supplement 6 Change 1 dated June 30 2005 or later approved revision. | | 20.02.2006 |
| 10015672 | AVCON INDUSTRIES, Inc. | EASA.IM.A.S.01650 | | ST01195W1 | LEARJET 24 LEARJET 35 LEARJET 36 | Learjet 20 Series RVSM Learjet 35/36 Series RVSM | | MDL 1078 Revision AGAFMS, Learjet Model 25D/F, Century III or Century III/Software WingAFMS, Learjet Model 24D, Dee Howard XR ModificationAFMS, Learjet Model 24EAFMS, Learjet Model 24D, Standard Wing with/without AMK 83-4AFMS, Learjet Model 25B/C, M (...) | As detailed in the AFM | 08.03.2007 |
| 10036439 | AVCON INDUSTRIES, Inc. | | | FAA STC SA00415W1 | 100, A100, A100A, A100C, B100 200, 200T, A200, B200, B200T 300, 300LW, B300 B90, C90, C90A, E90, H90 | Installation of Baggage Door. | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated-certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Installation in accordance with Avcon Industries Inc, Master DrawingList No 30500, Revision C, dated 27 March 1996.Instructions for Continued Airworthiness in accordance with Avconindustries Inc, Maintenance Manual Supplement Document No 00107025,Re (...) | None | 08.09.2011 |
| 10042976 | AVCRAFT AEROSPACE GMBH | EASA.IM.A.S.01241 | | FAA STC ST09465AC | DORNIER 328-100 | Modification of the cabin floor to be the same level through out thecabin, according to FAA STC No ST09465AC, reissued 18 April 2005 | | -Fairchild Aircraft Report R1530, Revision D dated 20 April1999-FAA STC ST09465AC, reissued 18 April 2005 | 1. According to the application as forwarded by FAA letter dated 27April 2005, the approval is limited to te aircraft S/N 30952. Furthermore the limitations as referenced in FAA STC ST09465AC, reissued 18 April 2005, apply.3. This STC is approved only (...) | 15.08.2005 |
| 10016623 | AVIATION CONSULTING & ENGINEERING | EASA.IM.A.S.03054 | | FAA STC SA01282W1 | CESSNA 525 | Installation of Replacement of Seat Cushions. | | - Cushions must be installed law, to Aviation Consultingand Engineering Solutions Master Data List Doc. No. A08190404 stamped #FAA Approved# from 21 Oct 2004, or later FAA approved Revisions.- Installation only approved on seat assemblies as listed in FAA STC SA01282W1. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 03.12.2008 |
| 10014067 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.01154 | REV. 2 | | BEECH 200 BEECH B200 | Installation of an Emergency Locator ELT 96 - 406 MhZ. | | Modification File AVD-MB200-23-046A. Supplemental AFM No. 1,3, & 6. | This STC is limited to Beechcraft 200 MSN BB 199, Beechcraft B200 MSN BB920 and Beechcraft 200 s/n BB439. | 15.03.2005 |
| 10014068 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.01155 | | | BEECH 200 BEECH B200 | Installation of 2 Mode S transponders MST 67A. | | Modification AVD-MB200-34-045A, Revision b, dated 20th May 2005 or later EASA-approved revision, Supplemental Maintenance Manual, Edition 4 Revision 2. | 1. This STC requires a pre-modification: 2 bottom transponder antennas.2. This STC is limited to Beech s/n BB199 and Beech 200 s/n BB439. | 28.07.2005 |
| 10014069 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.01157 | | | BEECH C90 BEECH E90 | Installation of an Emergency Locator ELT 96-406 MhZ. | | -Modification File AVD-MB90-23-042A- Supplemental AFM No 2 &3 | This STC is limited to Beechcraft C90 s/n LJ860 and Beechcraft E90 s/n LW251. | 28.07.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|----------------|----------------|----------------------|--|---|---|---|--|---------------|
| 10014070 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.01158 | | | BEECH C90 | Installation of Mode S Transponder MST 67A. | | Modification AVD-MB200-34-047A, revision a dated 20 June 2005 or later EASA approved revision. | 1. This STC requires a pre-modification: 2 bottom transponders antennas. 2. This Modification is limited to Beechcraft C90 s/n LJ860. | 22.07.2005 |
| 10014146 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.01265 | | | BEECH C90 | Replacement of Primus 40 weather radar by a Primus weather Radar. | | Modification File No. AVD-MB90-34-048A revision A of 4 July 2005 or later EASA approved Revision. | This STC is limited to Beechcraft C90 s/n LJ860. | 28.07.2005 |
| 10014212 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.01370 | | | BEECH 58 | Installation of emergency locator transmitter ELT 96-406 MHz in accordance with modification file AVD-MB58-23-044A. | | Airplane Flight Manual Supplement No. 1 / AvDef Beech 58, edition 1 dated 9 May 2005/Maintenance Manual Baron 55/58 edition 1, revision 1 dated 19 April 2006 | Modification limited to s/n TH-647The AFMS No. 1 / AvDef Beech 58 must be used with this modification. | 21.04.2006 |
| 10014584 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.02691 | REV. 1 | | FALCON 20 FAN JET | AVD-MDA20-95-054B: Installation of new age Alkan Pylon6075(76)C006 | | see Change Approval Sheet CAS AVD-MDA20-95-054B Iss. 1 dtd. 7.4.08 or later approved revisions;-AFM supplement SAFM FALC20 rev 13 dtd. 15.Sep.08 or later approved revisionsand associated technical documentation | 1. Limitations law SAFM FALC20 apply for operation with AVDMA20-95-054B2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification wi (...) | 24.11.2008 |
| 10014585 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.02692 | | | MYSTERE FALCON 20 | AVD-MDA20-95-055A: Installation of underwing stores | | see Change Approval Sheet CAS AVD-MDA20-95-055A Ed. 01 Rev.00 dtd.07.01.08 or later approved revision-AFM Supplement SAFM-FALC20 Rev. 14 dtd. 16.09.08- Instructions for Continued Airworthiness law Modification Sheet AVD-MDA20-95-055A Section B.8and associated technical documentation | 1. Limitations law AFM Supplement SAFM-FALC20 to DTM 918 apply for operation with POD installation law AVD-MDA20-95-055A2. Prior to installation of this modification the installer must determine that the interrelationship between this modification an (...) | 24.11.2008 |
| 10014586 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.02693 | | | MYSTERE FALCON 20-C5 MYSTERE FALCON 20-D5 MYSTERE FALCON 20-E5 | STRUCTURAL INSTALLATION OF A FUGRO GPS ANTENNA ON THE FUSELAGE | | Change Approval Sheet N AVD-MDA20-34-052A Issue 1 or later approved revisions and associated documents | Compatibility with previous approved modifications shall bedetermined by the installer. | 23.04.2007 |
| 10014587 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.02694 | | | MYSTERE FALCON 20 | AVD-MDA20-25-050A: Installation of consoles in cabin. | | see Change Approval Sheet CAS AVD-MDA20-25-050A Iss. 1 dtd.13.11.2007 or later approved revision and associated technical documentation. | | 15.11.2007 |
| 10014639 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.02785 | | | MYSTERE FALCON 20 | AVD-MDA20-49-053A: Additional Power Bus using APU generation, installation of 2 additional inverters. | | see Change Approval Sheet CAS AVD-MDA20-49-053A Iss. 0 dtd.18.4.06 or later approved revision -AFM supplement to DTM 918 - AFMS FA20 rev 12 dtd. 10.12.07 or later approved revision and associated technical documentation | Limitations law AFM supplement to DTM 918 AFMS FA20 apply for operation with additional inverters installed iaw AVD-MDA20-49-053A. | 10.01.2008 |
| 10014724 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.02901 | | | FALCON 10 | AVD-MDA10-34-060A: installation of EGPWS Mark VIII | | See change approval sheet CAS AVD-MDA10-34-060A Ed. 1 Rev. 2 dtd. 13 November 2007 (configuration per MS AVD-MDA10-34-060A).AFM Supplement to AFM DTM 721, Supavd1 EGPWS FA10 rev. 01 dated June 12, 2007.Associated technical documentation. | Limitations law AFM Supplement Supavd1 to DTM 721 apply foroperation with EGPWS installation iaw AVD-MDA10-34-060A.TGL12 complied with.Cert basis FAR 25 Amdt 20. | 26.11.2007 |
| 10014730 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.02909 | REV. 1 | | FALCON 20 | AVD-MDA20-95-057A: Activation of the SETHI 2 missionAVD-MDA20-95-054B: RAMHY Project - VHF/UHF Antenna | | see Change Approval Sheet CAS AVD-MDA20-95-057A Ed.1 dtd. 21.6.07 or later approved revisions:see Change Approval SheetCAS AVD-MDA20-95-073A issue 1 dtd. 21.11.08 and associated technical documentation- AFM supplement SAFM FALC20 rev 15 dtd. 17.09.0 (...) | 1. This STC ensures proper isolation of Payload (SETHI Mission Equipment) from / return to - civil configuration in case of inadvertent activation.2. Activated SETHI Mission Equipment is not approved by EASA.3. Limitations iaw AFM Supplement SAFM-FA (...) | 18.12.2008 |
| 10014760 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.02950 | | | BEECH 200 | Installation of two 8.33 kHz VHF | | None | This STC is limited to the S/N BB199 and BB920 | 04.04.2007 |
| 10014783 | AVIATION DEFENSE SERVICE (AVdef) | EASA.A.S.02981 | | | FALCON 20 | Enhanced Surveillance Mode S Transponder installation. | | AvDef Modification Sheet N° AVD-MDA-20-34-064A, Edition 01 Revision 03 dated 11 January 2008.Falcon 20 AFM Supplement AvDef N°1 Enhanced Mode S Installation issue 0 dated 8th January 2007.Or later approved revisions of these documents | Major change applicable to aircraft Fan Jet Falcon and Falcon 20 serial numbers 076, 103 and 280 equipped with:- Integrated Collins Proline IV avionics suite - DFS Elementary Mode S Transponder as per EASA.A.S.00203 STC - AvDef Minor change EASA.A. (...) | 16.01.2008 |
| 10037677 | AVIATION DEFENSE SERVICE (AVdef) | | | | FAN JET FALCON BASIC | Carriage of ARPEGE pylons only.The purpose of this approval is to clear the carriage of the ARPEGEpylons only configuration on Falcon 20 S/N 076. | The Certification Basis for the original product remains applicable tothis certificate/ approvalThe certificated noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval | Change Approval Sheet doc. N° CAS-AVD-MDA20-95-FOG1A Issue 3 dated14.12.2011Master Document List doc. N° MDL-AVD-FOG1A Issue 2 dated 14.12.2011Aircraft Flight Manual Supplement doc. N° SAFM-FALC20 Rev. 17or later revisions of the above listed documents approved by EASA | Restricted to Falcon 20 S/N 076 aircraft.As per AFM Supplement N° SAFM-FALC20 Rev. 17. | 14.12.2011 |
| 10040822 | AVIATION DEFENSE SERVICE (AVdef) | | | | FAN JET FALCON BASIC | Carriage of PIRATE pylons only.The purpose is to clear the carriage of a new configuration of underwingstores-<->- dedicated for ferry flights without the PIRATE pods. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS25 Amdt 10 §631 and FAR 25 §1529The certificated noise and/ or emissions levels of t (...) | Change Approval Sheet doc. N° CAS-AVD-MDA20-95-FOG5A Issue 01 dated26.07.2012.Master Document List doc. N° MDL-AVD-FOG5A Issue 2 dated 26.07.2012Aircraft Flight Manual Supplementdoc. N° SAFM-FALC20 Rev. 18;or later revisions of the abovelisted documents approved by EASA. | Restricted to Falcon 20 S/N 076 aircraft.As per AFM Supplement N° SAFM-FALC20 Rev. 18. | 27.07.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|-----------------------|--|--|--|--|--|---------------|
| 10034408 | AVIATION ENGINEERING AS | | | | AS 350 B2 & AS 350 B3 | Installation of Electromagnetic Ground Ice Thickness Sensor | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval FAR 27 at amendments as specified in Aviation Engineering AS COMPLIANCE CHECK LIST doc. (...) | Aviation Engineering AS Flight Manual Supplement AS350B2, B3 Post Modification Eurocopter SBT 25.00.62 - doc. No. 2587-02-353, rev. 0 d021.01.2011 for later revisions of the above listed documents approved by EASA Aviation Engineering AS Master Drawing (...) | AS350B2 and AS350B3 post mod Eurocopter SBT 25.00.62 | 01.04.2011 |
| 10042854 | AVIATION ENGINEERING AS | | | | AS365 N1, N2, N3 | Modification of HeliDyne DPL Litter P/N HD365-15-1036 Approved under STCLFVS 21E2.003 (Sweden). Rebuilding of the HeliDyne DPL Litter to facilitate Aerolite Platform P/N 007-0002633-501 (EMS Patient Platform). | The Certification Basis for the original product remains applicable to this certificate/ approval. | 2524-03-320 dated 25 April 2012 - Certification Plan 2524-03-322 Design Change Report | STC EASA.R.S.01072 to be installed | 20.12.2012 |
| 10015339 | AVIATION FABRICATORS / CENTRAL | EASA.IM.A.S.00389 | REV. 2 | | SEE ATTACHED EASA APPROVED TYP | Installation of aft Jump Seat Kits. | | Installation in accordance with Aviation Fabricators, Inc. Drawing List, as defined in the attached AML, or later approved revisions. | 1. Installation per the STC Drawing Lists defined in the AML is limited to aircraft with Beech factory installed oxygen system for the jump seat locations. Aircraft configured for more than 14 total occupants must be equipped with a 49 cubic foot or greater oxygen system. | 17.04.2008 |
| 10015427 | AVIATION FABRICATORS / CENTRAL | EASA.IM.A.S.01160 | | FAA STC ST00716W1 | CESSNA 500 CESSNA 501 CESSNA 550 CESSNA 551 CESSNA 560 CESSNA 650 CESSNA S550 | Installation of Shoulder Harness Kit | | Drawing List AF-288 of 29 August 2002 (Rev F)-Structural Report D-10314SA of 03 October 2000-Installation Instructions D-10252 (D), D-10314 (A) and D-10368-No change in the AFM-FAA STC ST00716W1 dated 02 November 1998 | | 22.07.2005 |
| 10015577 | AVIATION FABRICATORS / CENTRAL | EASA.IM.A.S.01488 | REV. 1 | FAA STC SA00622W1 | SEE EASA APPROVED MODEL LIST | Installation of shoulder harness kit | CAR 3 and FAR 23 | Installation in accordance with Aviation Fabricators, Inc FAA approved Drawing List AF-203 F, dated 6 October 2009 or later approved revision. | None | 11.12.2009 |
| 10015888 | AVIATION FABRICATORS / CENTRAL | EASA.IM.A.S.02007 | | FAA STC ST01262W1 | CESSNA 650 | Installation of a 1, 2 or 3 place side facing Divan | | Installation in accordance with Aviation Fabricators STC Data Package AF-263, Rev K, dated June 9, 2004, with an approval date of July 28, 2004, or later approved revision. | Applicable to aircraft serials 650-0023 and subsequent. This STC approves only the installation of the parts contained in the installation documents referenced above. The requirements for aisle width, passenger oxygen systems, seats and safety belts, a (...) | 09.01.2007 |
| 10015991 | AVIATION FABRICATORS / CENTRAL | EASA.IM.A.S.02145 | | FAA STC SA4147SW | HAWKER 1000 HAWKER 800 HAWKER 800 XP HAWKER HS 125 | Installation of one, two, three & four-place side facing seat. | | Installation of one, two, three & four-place side facing seat according to Custom Aircraft Conversions Drawing 01125 dated 12. June 1980, or later approved revision; or Aviation Fabricators STC Drawing List AF-177 dated 06. December 1994 or later approved revision | This STC grants no operational approval to this installation, that will need to be obtained from the State of registry for the particular aircraft involved as defined in JAR OPS1. | 30.10.2006 |
| 10016005 | AVIATION FABRICATORS / CENTRAL | EASA.IM.A.S.02168 | | FAA STC ST01139W1 | CESSNA 560 | Installation of a 39-inch or a 45-inch two-place divan or a 24-inch one-place divan per Aviation Fabricators STC Data Package, AF-342, Rev A, dated 12. June, 2003 and stamped "FAA Approved" with an approval date of October 22, 2003 or later approved revision | | STC Data Package, AF-342, Rev A, dated 12. June, 2003 and stamped "FAA Approved" with an approval date of October 22, 2003 or later approved revision | The STC covers only installation of parts contained in the Description section Upholstery, covers, aisle width, oxygen, seat belts are not covered | 13.09.2006 |
| 10015576 | AVIATION FABRICATORS, Inc. | EASA.IM.A.S.01487 | REV. 1 | FAA STC SA2629CE | 200, 200C, 200CT, 200T, 300 300LW, B200, B200C, B200CT B200GT, B200CGT B200T, B300, B300C | Installation of two place side-facing divan. | FAR 23 | Aviation Fabricators Drawing Number D10002B approved on 30 MAY 1990 or Aviation Fabricators STC data package AF-113, Rev. D or later approved revision. | 1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2) The (...) | 08.10.2009 |
| 10029817 | AVIATION FABRICATORS, Inc. | | REV. 1 | FAA STC ST01572W1 | 400 400A | Change to STC 10029817 Beechjet 400 Cabin Configurations, Dvan Installations. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation of four (4) model 400A OEM seats, one (1) two place divan, one (1) three place divan, one (1) aft toilet, one (1) set of aft pocket door dividers, one (1) set of forward pocket door dividers, or LED lighting, per the following: -1), Master (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 28.05.2013 |
| 10029869 | AVIATION FABRICATORS, Inc. | | | FAA STC NO. SA00961W1 | PA-31-325, PA-31-350 PA-31P, PA-31T, PA-31T1 PA-31T2, PA-31T3, PA-31 PA-42, PA-42-720, PA-42-1000 | Stretcher Installation per Aviation Fabricators, Inc. STC Drawing List, Doc. No. AF-204 | Design: Non-significant change The Certification Basis of the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | FAA approved STC Drawing List, Doc. No. AF-204, Rev. (D), dated 20 Jan 2010, or later FAA approved revision Approved Instructions for Continued Airworthiness, Doc. No. AF-489, Rev.(A), dated 07-Jan-2010, or later approved revision | The installer is responsible for ensuring that the installation complies with any supplemental oxygen requirements as required for operation | 27.04.2010 |
| 10031917 | AVIATION FABRICATORS, Inc. | | | FAA STC SA2818SW | 441 | Installation of 1, 2 or 3 place side seating divans in accordance with FAA STC SA2818SW | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | STC Drawing list, AF-294, Rev. E, dated September 1, 2010 Installation instructions: D-10273, Rev. C, dated September 1, 2010 Installation location instruction (1 place, 2 place, 3 place divan): AF-497, AF-498, AF-499 accordingly, no Rev., all dated Mar (...) | none | 22.09.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|----------------------|--|--|---|--|--|---------------|
| 10032446 | AVIATION FABRICATORS, Inc. | | | FAA STC SA2774CE | 1. 200, 200C, 200CT, 200T 1. A200, A200C, A200CT, B200 1. B200CT, B200T, 300, 300LW 2. B200C, B200CGT, B200GT 2. B300, B300C | Installation of Passenger Seats with Hinged Backrest. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Aviation Fabricators Drawing List Number AF-151 FAA Approved 12 November 1991 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | none | 08.11.2010 |
| 10032903 | AVIATION FABRICATORS, Inc. | | | FAA STC REF SA4157SW | SEE EASA APPROVED MODEL LIST | Installation of one, two, or three place side facing seat (frame only). | FAR 23, plus Amendments listed in the EASA or FAA TCDS, as appropriate to the Model. The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Aviation Fabricators Drawing Ref B6912 dated 7/4/80, or drawing list AF-291, no revision, dated August 14, 2002 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None | 07.12.2010 |
| 10046164 | AVIATION FABRICATORS, Inc. | | | FAA STC SA2847CE | CESSNA 208, 208B | Stretcher assembly with restraints and portable Ferno unit can be installed on existing cabin seat track on either side of the aircraft between FS 158.00 and FS 238.00 in accordance with FAA STC SA2847CE. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Instructions of Continued Airworthiness AF-507, Rev. B, dated October 22, 2012; Drawing List AF-121, Rev. E, FAA approved February 19, 2013; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (...) | This approval is applicable only to Cessna 208 and 208B cargo airplanes that have a right hand cabin entry door or the Cessna kits installed to permit opening the left hand cargo door from inside the cabin. The installer must determine whether this des (...) | 23.08.2013 |
| 10015440 | AVIATION FLEET SOLUTIONS, LLC | EASA.IM.A.S.01180 | REV 2 | FAA STC ST01649SE | MD-81, MD-82, DM-83, MD-87, MD-88 | Supplemental Operating Configurations MD-80. Addition of engine and part intermix. | | MDL AA 1781 (Rev. H), AFM Supplement | | 26.10.2007 |
| 10016689 | AVIATION FLEET SOLUTIONS, LLC | EASA.IM.E.S.01001 | REV 1 | FAA STC SE01650SE | PW JT8D-217, PW JT8D-217A PW JT8D-217C, PW JT8D-219. | Supplemental Noise Configuration: noise addition of normal TO Thrust + Automatic Reserve Thrust System and Landing Flap 28. The previous FAA STC SE01650SE dated 21.03.2006 was approved under EASA approval EASA.IM.E.S.01001 dated 6 June 2006. | | FAA approved AFM Supplement AFMS-MD80-1, Rev. A, dated 3 August 2006 for Airplane models MD-81/-82/-83/-88. FAA approved AFM Supplement AFMS-MD87-3, Rev. A, dated 3 August 2006 for Airplane models MD-87. FAA approved AFM Supplement AFMS-MD80-9 (normal T (...) | 1. See above AFM Supplements. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft (...) | 12.02.2007 |
| 10015323 | AVIATION PARTNERS BOEING | EASA.IM.A.S.00279 | REV. 3 | FAA STC ST00830SE | BOEING 737-700 BOEING 737-800 BOEING 737-900 | Installation of Aviation Partners Boeing Winglets in accordance with the following documents: - APB Document List AP37.7-E620 Rev. A (737-700)- APB Document List AP37.1-E620 Rev. A (737-700/GW)- APB Document List AP37.6-E620 Rev. A (737-800)- APB D (...) | | | 1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 2. If transfer of this Certificate is requested under Part 21A.116 the STC w (...) | 30.06.2008 |
| 10015659 | AVIATION PARTNERS BOEING | EASA.IM.A.S.01636 | REV. 5 | FAA STC ST01518SE | B757-200, -300 | Install of Aviation Partners Boeing Winglets on B757-200 and B757-300 Aeroplanes | Model 757-200 Parts, components, systems or areas not modified or affected by the installation of the APB winglet retrofit kit will comply through the same JAR amendments and exceptions as the EASA certification basis for the unmodified 757-200 as defin (...) | B757-200 Aviation Partners Boeing Approved Model List (AML) AP57.2-AML1518 Issued date 13th March 2009 Aircraft Flight Manual Supplement AP57.2-0601, Revision 4, 11th July 2007 B757-300 Aviation Partners Boeing Approved Model List (AML) AP57.3-AML1518, 16t (...) | The Partial Wing Retrofit Installation (300-1 Kit) is approved for Rolls Royce RB211-535 powered a/c. | 11.04.2011 |
| 10016394 | AVIATION PARTNERS BOEING | EASA.IM.A.S.02723 | REV. 3 | FAA STC ST01920SE | BOEING 767-300, -300F | Installation of Aviation Partners Boeing Winglets on Boeing B767-300 and -300F a/c. | The type certification basis for the B767-300/-300F series aeroplanes shown on the TCDS listed above for parts not changed or affected by this STC. For parts changed or affected by this STC the Certification Basis is EASA Certification Specification (...) | Model, Description, Approved Model List (AML Document), Model, Description, Approved Model List (AML Document), 767-300, Partial Wing Retrofit Installation (767-300-1 Kit), AP67.3-AML1920 Rev IP, 15th October 2008 B767-300, Partial Wing Retrofit Install (...) | Operational/ Performance limitations are defined within the AFM supplement. This STC is approved for a/c equipped with General Electric CF6-80C2 series engines or Pratt and Whitney PW 4000 series engines with no engine strut vibration isolators installed (...) | 07.01.2011 |
| 10029056 | AVIATION PARTNERS BOEING | EASA.IM.A.S.02838 | | FAA STC ST01219SE | BOEING 737-300, 737-500 | | | | | 15.07.2009 |
| 10015403 | AVIATION PARTNERS, INC. | EASA.IM.A.S.01113 | | FAA STC ST01411SE | BAE 125 SERIES 800A BAE 125 SERIES 800B HAWKER 800 XP | Installation of Winglets, associated changes in wing and Aileron structures and related systems changes on Raytheon Aircraft Company Model Bae 125 Series 800A, 800B and Hawker 800XP Aircraft, in accordance with FAA STC ST01411SE | | Aviation Partners Inc Master Drawing List ref APH-0605, Revision A, dated May 12, 2004 for Models 800A and B, and ref APH-0610, Revision A, dated November 23, 2004 for the Model 800XP, or later approved revisions Aircraft modified in accordance with (...) | Installation of this STC on Model Hawker 800XP is limited to Aircraft equipped with the Honeywell DFZ-800 (SPZ-800 avionics) autopilot, the Collins 85/86 (FCC-86E avionics) auto pilot and Collins Pro line 21 (FGC-300 avionics) autopilot, for Categor (...) | 26.10.2005 |
| 10028844 | AVIATION PARTNERS, INC. | EASA.IM.A.S.02043 | | FAA STC ST01987SE | DASSAULT FALCON 2000 DASSAULT FALCON 2000EX SERIES | | | | | 08.05.2009 |
| 10038665 | AVIATION PARTNERS, INC. | | REV. 2 | FAA STC ST02188SE | MF900, F900EX | F900 aircraft fitted with winglets per EASA STC 10038665 Steep Approach Landing Approval | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): B-07 - Steep Approach The requirements for environmental protection and the associated certified n (...) | FAA Certification Flight Test Plan ref. ATEC-12022 Rev A dated August 14, 2012 AFM referenced APF9-0601_Code_9001 Revision 1 April 15, 2013 Master Drawing List Revision D, December 6, 2011 or later revisions of the above listed documents approved by EASA | Applicable to all Dassault Aviation MF 900 and F900EX aircraft fitted with EASA STC 10038665 Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification (...) | 24.04.2013 |
| 10015248 | AVIATION SERVICE A.S. | EASA.A.S.03687 | | | LETECKE ZAVODY L 410 UVP-E | Installation of EGPWS Mark VI. | | # Flight Manual Supplements:- No: FM 34 41/01 001 (Multi Functional Display KMD 540), dated Nov 11/08- No: FM 34 43/01 001 (EGPWS Mark VI), dated Nov 11/08- No: FM 34 55/01 001 (Radar Altimeter KRA 405B), dated Nov 11/08# Engineering Bulletin:- No: EB 166, dated 11.11.2008 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 24.11.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------|------------------|----------------|----------------------|---|--|--|---|--|---------------|
| 10029324 | AVIATION SERVICE A.S. | P-EASA.A.S.03854 | | | L 200 A, L 200 D | Installation of Aspen EDF-1000 "PRO" and EFD-1000 "PILOT", GNS 430W, GTX330, KR 87, KN 62A and ELT ME406 systems. | CS-23, Amdt. 1, dated 12 February 2009 including JAA Interim Policies No: INT/POL/23/1 (HIRF) and INT/POL/23/3 (IEL). | L 200 D Flight Manual Supplements No... , FM 34 20/01 001 Primary Flight System EFD1000 Pro, dated Nov 27/09... , FM 34 20/01 002 Primary Flight System EFD1000 Pilot, dated Nov 27/09... , FM 34 50/01 001 GPS/WAAS NAV COMM GNS 430W, dated Nov 27/09... , FM 34 (...) | None. | 18.03.2010 |
| 10030602 | AVIATION SERVICE A.S. | | | | L410 UVP-E, L410 UVP-E9, L410 UVP-E20 | TCAS II CAS67A, Mode S transponder MST67A, and Air Data Display Unit AD32 installation to comply with regulation requirements. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval-FAR Part 23, Amdt. 49 | -FM34 40/01 001., Flight Manual Supplement CAS 67A and MST 67A-FM34 40/01 002., Doplněk Letové příručky CAS 67A a MST 67A-FM34 10/01 001., Flight Manual Supplement AD 32-FM34 10/01 002 Doplněk Letové příručky AD 32 | None | 24.06.2010 |
| 10031903 | AVIATION SERVICE A.S. | | | | 407 | Installation of alternate static port for VFR night operation | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | FM 30 30/01 001 revision 0 dated 24 August 2010 - Flight Manual Supplement - alternate static port or later revisions of the above listed documents approved by EASA AS 228 001P issue 0 dated 14 July 2010 - Instructions for fabrication, installation and (...) | None | 21.09.2010 |
| 10033022 | AVIATION SERVICE A.S. | | | | AN-26B | An-26 Aircraft Upgrade; TCAS II CAS-67A, Mode S Transpon. MST-67A and KT73, EGPWS Mk-VI, NAV/COMM KX165A, GPS155XL, Audiopanel KMA28, Wx Radar ART2000, MFD KMD540, ADF KR87, DME KN63, Air Data Display AD32, RAIt KRA405B, Airspeed Ind. LUN110, ELT C406-2 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval-CRI A-100The requirements for environmental protection and the associated certified n (...) | CP226 and CHH226 | None | 15.12.2010 |
| 10045148 | AVIATION SERVICE A.S. | | | | 429 | Dual-Color Anti-Collision Light Installation for Bell 429 rotorcraft. The change is described in Aviation Service Engineering Bulletin EB-241-0 dated February 19, 2013. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Aviation Service Flight Manual Supplement FM_33_40/03_001 Rev.5 Dated April 29, 2013; or later revisions of the above listed documents approved by EASA. Engineering Bulletin EB-241 Dated 19/02/2013. Maintenance Manual Supplement MM 33_40/03_001 Rev.0 dated February 18, 2013. | The EASA approval of this change does not constitute operational approval. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 03.06.2013 |
| 10045150 | AVIATION SERVICE A.S. | | | | BELL 429 | Installation of WESCAM MX-10 on Meesker Nose Mount AFM-NM429-1, and Skyforce Moving Map System OB3000/3300, Aviation Service Operator workstation including F+E seat H110SU with Skyquest monitor AVDU 5008, and Convergent Nanoflash digital video recorder (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Aviation Service Moving Map Flight Manual Supplement FM_34_40/03_001 Rev.5, dated June 3, 2013.-Aviation Service Radio Provision Flight Manual Supplement FM_34_40/03_002 Rev. 2, dated April 17, 2013.-Aviation Service Wescam Camera Flight Manual Suppl (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 04.06.2013 |
| 10017094 | AVIATION SUPPORT GmbH | EASA.R.S.01006 | | | EUROCOPTER DEUTSCHLAND BO-105 | Installation of a Gas Detection System CHARM with a laser scanner | | Flight Manual Supplement "Ergänzung zum Flughandbuch FMS-ASE-90411, Rev. -" or later approved revisions. Installation Instructions "Installationsanleitung REP-ASE-90411-12, Rev. 3" dated 29-11-2006 or later approved revisions + "Installationsanlei (...) | | 13.02.2007 |
| 10017159 | AVIATION SUPPORT GmbH | EASA.R.S.01267 | | | SCHWEIZER AIRCRAFT 269A SCHWEIZER AIRCRAFT 269B SCHWEIZER AIRCRAFT 269C SCHWEIZER AIRCRAFT 269C-1 SCHWEIZER AIRCRAFT 269D | Installation of Clipboard/Tablet-PC rack | | Flight Manual Supplement: "Ergänzung zum Flughandbuch FMS-ASE-90416" Installation Instructions: "Montageanleitung ASE-90416/6" and "Installationsanleitung ASE-90416/7" | -- | 30.01.2007 |
| 10027152 | AVIATION SUPPORT GmbH | P-EASA.R.S.01558 | | | MD 600N | Installation of HD Camera System and right side dual utility mount MD600N, as per Master Document List Ref MDL-ASE-90819, dated 08.09.2009 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged an (...) | Instruction for Continued Airworthiness Ref. ICA-ASE-90819, Rev 0.3, dated 08.09.2009 - Flight Manual Supplement Ref FMS-ASE-90819, Rev 0.4, dated 08.09.2009 or later revisions of the above listed documents approved by EASA | Approval limited to S/N RN073 | 10.09.2009 |
| 10027605 | AVIATION SUPPORT GmbH | P-EASA.R.S.01336 | | | AS350 B, D, B1, B2, BA, BB, B3 | Installation of a 3D laser measurement system in an external cargo pod. | FAR 27 The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | ICA-ASE-90603 Revision - Instructions for Continued Airworthiness; -MDL-ASE-90603 Revision - Master Document List (Liste der gültigen Dokumente); -FMS-ASE-90603 Revision - or later EASA approved revision - Flight Manual Supplement" or later revisions of (...) | None | 13.10.2009 |
| 10030519 | AVIATION SUPPORT GmbH | | | | AS350B, D, B1, B2, BA, BB, B3 | Cargo Mirror Set Installation | As per EASA TCDS.R.008 Issue 04, 23 November 2009 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions level (...) | Definition and installation Installation Manual REP-ASE-90824-4, Rev. -, dtd 19.11.2009 Inspection, maintenance, and operation instructions for Continued Airworthiness ICA-ASE-90824, Rev. -, dtd 25.03.2010 Rotorcraft Flight Manual Supplement FMS-ASE-90824 (...) | None | 22.06.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|-------------------|----------------|----------------------|--|--|--|--|--|---------------|
| 10032031 | AVIATION SUPPORT GmbH | | REV. 1 | | MD900 | Installation of an O2 Rack | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CS-27 Amendment 2 for the change The requirements for environmental protection and the ass (...) | RFM Supplement; FMS-ASE-91 001 or later revisions of the above listed documents approved by EASA Master drawing list; DL-ASE-91001 (22.3.10) Compliance Report; CR-ASE-91001 (6.9.10) ICA; I CA-ASE-91 001 (30.8.10) | This change is restricted to Serial number 900-00105 only. | 08.12.2010 |
| 10033427 | AVIATION SUPPORT GmbH | P-EASA.R.S.01335 | | | MD900 | Installation of swiveling speaker system MD900 (ALS-ausfahrbares Lautsprecher-System MD 900) | JAR 27 | - EASA approved Flight Manual Supplement FMS-ASE-90304, Rev. -- Maintenance Manual Supplement MMS-ASE-90304, Rev. - - Master Document List MDL-ASE-90304, Rev. -. | none | 23.06.2009 |
| 10033443 | AVIATION SUPPORT GmbH | P-EASA.R.S.01513 | | | EC135 | Installation of High Definition (HD) Broadcasting System | CS-27 | 1. Flight Manual Supplement FMS-ASE-90818 High Definition Broadcast System (HDBS) Eurocopter EC 135 Series 2. MAINTENANCE MANUAL SUPPLEMENT MMS-ASE-90818 HD Broadcasting System EC135 series 3. Installationsanleitung REP-ASE-90818-4 HD Broadcasting System (...) | With HDBS camera system installed: VMC DAY ONLY; Category "A" operation prohibited. Antenna Actuator: Minimum height to deploy antenna: 50ft AGL Maximum flight speed to deploy antenna is 75 KIAS until antenna is fully deployed. | 06.07.2009 |
| 10033667 | AVIATION SUPPORT GmbH | | | | MD900 | Installation of an Equipment Rack | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CS-27 Amdt. 2 The requirements for environmental protection and the associated certificate (...) | Master drawing list; MDL-ASE-90906 Rev 0-1 Compliance Report CR-ASE-90906 Rev 0-1 ICA; ICA-ASE-90906 Rev 0-2 RFM Supplement; FMS-ASE-90906 Rev 0-2 or later revisions of the above listed documents approved by EASA | None | 02.02.2011 |
| 10033670 | AVIATION SUPPORT GmbH | | | | 1. AB 206 A & AB 206 B 2. 206A & 206B | Installation of ALMA (Airborne Laser Methane Assessment) - Gas Detection System. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CS27 Change 2 The requirements for environmental protection and the associated certificate (...) | MDL-ASE-90902 revision 2-1 dated 2 February 2011 - Master Document List; CR-ASE-90902 revision - dated 1 September 2010 - Compliance Checklist; FMS-ASE-90902 revision 0-1 dated 2 February 2011 - Flight Manual Supplement; ICA-ASE-90902 revision 0.1 dated (...) | none | 02.02.2011 |
| 10034980 | AVIATION SUPPORT GmbH | | REV. 1 | | SA 365 N | Title: SA 365 EMS Equipment. Description: Installation of an EMS Equipment fixed at the seat and cargo attachment points of the passenger compartment. The System is installed at the location of the first and second seat row. The aft seat row remains for 1 (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight manual Supplement, EMS Equipment, SA365N, ASE-91012, FMS-ASE-91012, Revision 0-3 dated 5th May 2011 (<->) or later revisions of the above listed documents approved by EASA. Liste der gültigen Dokumente (Master Document List), ASE-91012, EMSEqu (...) | This STC is applicable for the Serial Number 6100. The use of the electrical components of the medical equipment is limited to VFR Operations. The use of a hoist is not possible if the EMS Equipment is installed. | 24.05.2011 |
| 10039242 | AVIATION SUPPORT GmbH | 0010012010-001 | | FOCA STC 25-70-012 | EC135 T2 | Installation of an external Ski Basket Equipment P/N AV 115.00 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Rotorcraft Flight Manual Supplement Doc. Number FMS SUP.11-110- Instructions for Continued Airworthiness Doc Number AZ-ICA AZE-Basket EC 135-AV115.00 Edition 1 Rev 0, dated 12.10.2004 or later revisions of the above listed documents approved by EASA- (...) | None | 19.04.2012 |
| 10046138 | AVIATION SUPPORT GmbH | | REV. 1 | | BO105 D (RESTRICTED TO VARIANTS BO105 DBS-4 + DBS-5) BO105 S (RESTRICTED TO VARIANTS BO105 CBS-4 + CBS-5) | Title: Camera Mount for BO105 Variants. Description: Installation of a camera mounting device for fitting Cineflex HD series (and equivalent) camera systems to the LH side of the Eurocopter BO105 landing gear. An existing step utility mount that already (...) | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS27 Amdt. 2: CS27.1, CS27.21, CS27.25, CS27.27, CS27.29, CS27.31, CS27.45, CS27.49, CS27.51, CS27.65, CS27.67, CS27.71, CS (...) | Flight Manual Supplement, Camera Mount BO105 (CBS-4, CBS-5, DBS-4, DBS-5), ASE-91209, FMS-ASE-91209, Rev-0-3 dated 2nd July 2013 or later revisions of the above listed documents approved by EASA. - Projektbeschreibung ASE-91209, Camera Mount BO105, PB- (...) | None | 04.09.2013 |
| 10015418 | AVIATION TECHNICAL SERVICES, Inc. | EASA.IM.A.S.01141 | | FAA STC ST00988SE-D | BOEING 757-200 | Installation of an Interior Arrangement, Avionics & Ozone Converters. | | FAA STC No ST00988SE-D, Goodrich Master Drawing List 2004-005 Revision A, dated May 4 2004 or later approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.04.2005 |
| 10015543 | AVIATION TECHNICAL SERVICES, Inc. | EASA.IM.A.S.01428 | | ST00299SE-D | BOEING 737-300 BOEING 737-400 BOEING 737-500 | Modification of an 87MCU FMC Tray to accept a 4MCU FMC LRU according to FAA STC ST00299SE-D | | BF Goodrich Master Drawing List Ref. 98-065, Rev. C, dated 13 August 2001 or later FAA approved revision FAA STC ST00299SE-D, Date issued 5 July 2005 | According to FAA STC ST00299SE-D | 29.09.2005 |
| 10016147 | AVIATION TECHNICAL SERVICES, Inc. | EASA.IM.A.S.02348 | | FAA STC SA7030NM-D | BOEING 737-300 | Installation of Allied Signal p/n 967-0102-001 Digital Flight Data Acquisition Unit (DFDAU) and necessary sensors and wiring to record Twenty Two (22) Parameters of data per Tranco Incorporated Master Drawing List 95-180 Sheet 1 Revision A dated January 12, 1996 or later FAA-DAS approved revision. | | Tranco Incorporated Master Drawing List 95-180 Sheet 1 Revision A, dated January 12, 1996 or later EASA approved revision. | 1. Approval of this change in Type Design applies to the above model aircraft only. This approval should not be extended to other aircraft of this model on which other previously approved modifications are incorporated unless it is determined tha (...) | 27.02.2007 |
| 10016176 | AVIATION TECHNICAL SERVICES, Inc. | EASA.IM.A.S.02405 | | FAA STC SA5115NM | BOEING 737-300 | Installation of a Collins Traffic Alert and Collision Avoidance System (TCAS II) Transmitter/Receiver with Honeywell TA/RA/IVSI's. | | # FAA STC SA5115NM# ATS, Inc. Top Drawing 90-114, Revision F, dated January 05, 1995# Airplane Flight Manual Supplement D2000-163 approved February 22, 2001- or later revisions of the above documents approved by EASA in accordance with EASA (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 22.04.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|---------------------|----------------|----------------------|--|---|--|---|--|---------------|
| 10016177 | AVIATION TECHNICAL SERVICES, Inc. | EASA.IM.A.S.02406 | | US FAA STC SA4978NM | BOEING 737-300 | Installation of Traffic Alert and Collision Avoidance System (TCAS) provisions | | Tramco Top Drawing 90-078 Revision A dated August 20, 1990, or later approved revision | This approval is applicable only to the electrical provisions for TCAS and does not approve the actual TCAS installation or any TCAS equipment. | 12.07.2007 |
| 10029541 | AVIATION TECHNICAL SERVICES, Inc. | P-EASA.IM.A.S.02553 | | FAA STC ST00051SE | BOEING 737-400 | Installation of a Collins Traffic Alert and Collision Avoidance System(TCAS II) Transmitter/Receiver per Goodrich ATS, Inc Top Drawing 93046Revision F dated 05 January 2001. | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Goodrich ATS, Inc Top Drawing 93046 Revision F dated 05 January 2001- Airplane Flight Manual Supplement: Document D2000-181 Original/Revision Approved by FAA on 22 February 2001, or later revisions of the above listed documents approved by EASA inacc (...) | None. | 30.03.2010 |
| 10029850 | AVIATION TECHNICAL SERVICES, Inc. | P-EASA.IM.A.S.02613 | | FAA STC ST00631SE-D | B737-800 | Interior Reconfiguration - Installation of an Interior Arrangement, Entertainment System Components and Avionics Components. | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Goodrich Master Data List 2001-335 Rev. E dated 21/10/2002 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | Not Applicable | 26.04.2010 |
| 10014001 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.01038 | | | BOEING 757-200 | Installation of Enhanced Mode S Transponders | | -Aviation Traders Ltd Definition Document DD/021/9162 Issue2 or later approved revision -Aviation Traders Ltd AFMSupplement Number 31 Issue 1 or later approved revision | | 17.03.2005 |
| 10014002 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.01039 | | | BOEING 737-700 | Installation of Enhanced Mode S Transponders | | -Aviation Traders Ltd Definition Document DD/022/9218 Issue2 or later approved revision -Aviation Traders Ltd AFMSupplement Number 29 Issue 1 or later approved revision | | 17.03.2005 |
| 10014003 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.01040 | | | BOEING 737-300 | Installation of Enhanced Mode S Transponders | | -Aviation Traders Ltd Definition Document DD/023/9219/ Issue 2 or later approved revision -Aviation Traders Ltd AFM Supplement Number 30 Issue 1 or later approved revision | | 17.03.2005 |
| 10014163 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.01295 | | | | Installation of Teledyne Flight Data Acquisition Unit P/N 2233000-811 to provide parameters to the FDR | | | | 22.11.2005 |
| 10014175 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.01310 | | | COMMANDER AIRCRAFT 114B | Wing auxiliary Spar Repair | | Modification Summary Sheet - ALT/COM/M001 | | 13.10.2005 |
| 10014254 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.01479 | | | BOEING 707-300 | Avionics Upgrade | | 1. ATL Modification ATL/707/M127.2. Compliance Plan CP/032/9501 Issue 6 or later EASA approved revision.3. Flight Manual Supplement ATL AFM Supplement Number 34 Issue 1 or later EASA approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 15.10.2008 |
| 10014322 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.01575 | REV. 1 | | AIRBUS A300 B4 | Introduction of Enhanced Mode S Transponder | | - Aviation Traders Limited Modification 9613,- ATL AFM Supplement 35 Issue 3, or later approved revision, SSR Mode S Elementary & Enhanced Surveillance (Without LHAS-STC ST01149LA Installed),- ATL AFM Supplement 47 Issue 1, or later approve (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 14.09.2006 |
| 10014323 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.01576 | | | BOEING 727 SERIES | Introduction of Terrain Awareness & Warning System (TAWS) | | -Aviation Traders Modification ATL/727/M165 Issue 3-Aviation Traders AFM Supplement No 37 | | 25.01.2006 |
| 10014358 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.01623 | | | BOEING 757-200 | Introduction of Enhanced Mode S Transponder | | Aviation Traders Modification 9673 ATL AFM Supplement 36 | | 18.01.2006 |
| 10014372 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.01646 | | | BOEING 747-200 | Introduction of a Litton 72R INS in a Boeing 747-200 series aircraft | | 1. Compliance Plan Inertial Navigation System - Litton LTN-72R, ATL document CP/037/9732, Issue 1.2. AFL AFM Supplement Number 39 Inertial Navigation System, ATL document ATL/747/M032, Issue 13. Engineering Order Inertial Navigation System - Litton LT (...) | A copy of this Certificate and the EASA Approved Flight Manual Supplement (AFMS) must be maintained as part of the permanent records for the modified aircraft. | 19.06.2006 |
| 10014499 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.02564 | REV. 1 | | PARTENAVIA P68C | Camera Port Installation | | -- | -- | 13.12.2006 |
| 10014594 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.02706 | | | BAE 146 | Honeywell TCAS II | | ATL Modification ATL/146/M058ATL AFM Supplement Number 43 Issue 1 or later approved revision | | 05.07.2006 |
| 10014595 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.02712 | | | | Installation of Honeywell MKVII EGPWS. | | Airplane Flight Manual Supplement No. 44 Issue 1, dated 7th September 2006.ATL Definition Document No. DD/038/10031 dated 7th September 2006. | Compatibility of this design with previously approved modifications must be determined by the installer. | 31.05.2007 |
| 10014633 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.02775 | | | BAE 146 | Installation of dual AlliedSignal GNS-XLS Flight Management System | | Aviation Traders Modification ATL/146/M063Aviation Traders AFM Supplement Number 49 | Restricted to VH-JJ ASN E2113. | 10.05.2007 |
| 10014660 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.02817 | | | BOEING 737-300 BOEING 737-400 BOEING 737-500 | Mode 'S' Transponder System for Elementary and Enhanced Surveillance/Revision to EASA.A.S.02617 (B737-400 added to existing STC) | i.a.w. TCDS | - Modification Summary ATL/737/M556 Issue B, dated 1st May 2009.- ATL AFM Supplement Number 48 Issue 1. #SSR Mode #S/Elementary and Enhanced Surveillance, for B-737-300 series.EASA Form 96 Post-TC Technical Visa Page 3/7/Issue 2, 25.02.2008-ATL AFM (...) | In accordance with Paragraphs 3, 9, 10, 13 & 16 of the Modification Summary ATL/737/M556 Issue 1, dated 9th October 2006/Compatibility of this design change with other previously approved modifications must be determined by the installer. | 20.07.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|------------------|----------------|----------------------|---|--|--|---|---|---------------|
| 10014707 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.02879 | | | BOEING 747-400 SERIES | Installation of a Bendix / King CAS-81 TCAS II System in B747-400 in accordance with Master Drawing List Drawing No. 699-50021, initial release and Dual Bendix / King TRA-67 Mode S Transponder in accordance with Master Drawing List, Drawing 699-5002 (...) | | ATL AFM Nr. 54 Issue 1ATL Airplane Flight Manual SupplementNr. 54 to Boeing AFM D6-U10003. | Prior to installation of the STC Boeing SB 747-34-2309 must be installed. | 22.03.2007 |
| 10014708 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.02880 | | | BOEING 747-400 SERIES | RTA-44D VHF Data Radio System. | | None | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph 4: Specification Aircraft/Modification. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 01.02.2007 |
| 10014709 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.02881 | | | BOEING 747-400 | Certification of Collins Data Link ACARS Installation in the B747-400 | | MDL 94-24Page Avjet Master Drawing List for installation of Collins ACARS Part No 622-9717-00X on B747-400 series aircraft.Rockwell Collins drawing index dated 28th Sept 1994 superceding Page Avjet Master Drawing List MDL 94-24 of the samenummer | MDL 94-24Page Avjet Master Drawing List for installation of Collins ACARS Part No 622-9717-00X on B747-400 series aircraft. | 31.01.2007 |
| 10014734 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.02917 | | | BOEING 757-200 | 238 PAX LOPA Installation | | Aviation Traders Ltd. ATL Modification ATL/757/152 at Issue 1 dated 16 March 2007 or later approved revision.Aviation Traders Ltd. Master Document List ATL/757/MDL054 at Issue 1 dated 20 March 2007 or later approved revision. | | 21.03.2007 |
| 10014757 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.02945 | | | BOEING 757 | 201PAX LOPA Installation | | Modification ATL/757/158 | | 03.04.2007 |
| 10014887 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.03116 | | | BAE 146/AVRO 146 SERIES | 95 Passenger LOPA Installation. | | ATL/146/7067 Issue 2 | | 05.09.2007 |
| 10014940 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.03198 | | | | 148 Pax LOPA Installation | | - Modification Summary ATL/737/605 Issue 2 - DRN10512-001 Issue 2 | Compatibility of this design change with other previously approved modifications must be determined by the installer. | 05.09.2008 |
| 10014942 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.03201 | | | BOEING 757-208 | 219 PAX LOPA Installation. | | ATL Modification Summary Sheet ATL/757/180, issue 1, 02/08/2007. | This STC is limited to aircraft s/n 26270. | 21.08.2007 |
| 10014963 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.03234 | | | BOMBARDIER CL-600-2B16 | Introduction of Honeywell Mark V EGPWS. | | Aviation Traders Ltd Modification ATL/600/002ATL AFM Supplement No63 Issue 1 or later EASA approved revision | Aviation Traders Ltd Modification ATL/600/003 (EASA.A.S.03244) must be embodied prior to, or concurrently with, Modification ATL/600/002.Prior to installation of this modificationthe installer must determine that the interrelationship between this m (...) | 18.07.2008 |
| 10014967 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.03244 | | | BOMBARDIER CL-600-2B16 | EFIS, GPS approach scaling and EGPWS display | | Aviation Traders Ltd Modification ATL/600/003 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 18.07.2008 |
| 10014994 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.03291 | REV. 2 | | B757-200 | Passenger/ Cargo Combi Configuration | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificateFor changed areas Certification Basis is CS 25 amendment 2; includingthe following:CRI D-01: Spec (...) | ATL Master Document List: MDL-11630-00001, Issue 3, 04/02/2011ATL Modification Summary Sheet: MOD-11630-00001, Issue 2,04/02/2011or later revisions of the above listed documents approved by EASA | This STC is limited to a/c serial number 25621 onlyATL Minor Modification MOD/11630/0003 must be installed concurrentlywith this STC. | 07.02.2011 |
| 10015097 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.03436 | REV. 1 | | A310-203 A310-203C A310-204 A310-221 A310-222 A310-304 A310-308 A310-322 A310-324 A310-325 | Bluesky D1000 Satcom installationThe Supplemental Type Certificate EASA.A.S.03436 embodied provisions for a further configuration installing BlueSky Network Satcom System D1000withan Intercom Integration option.This revision 01 approves the activation of the installed device. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CI.DGAC-F TCDS TCDS n° 145CRI H-01 : ICA on EWISThe requirements for environmental pr (...) | Master Drawing List, ref. MDL/A31/017, Issue 6Aircraft flight Manual Supplement, ref. P1361-AFM-00001, Issue 1or laterrevisions of the above listed documents approved by EASA. | None | 21.03.2011 |
| 10015253 | AVIATION TRADERS LIMITED - ATL | EASA.A.S.03698 | | | BOEING 747-100 BOEING 747-200 | Installation of a Solid State Cockpit Voice Recorder and Control Unit in Boeing 747-100 and #200 series aircraft per ATL Master Document List MDL-11053-00001 at Issue 1 or later approved revisions. | | None. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 21.03.2009 |
| 10017153 | AVIATION TRADERS LIMITED - ATL | EASA.R.S.01256 | | | SCHWEIZER AIRCRAFT 269D | Introduction of EAC-333-000 Airborne Camera System | | Aviation Traders Modification ATL/333/M003 & Definition Document ref DD/001/9700 dated 14 March 2006Aviation Traders RFMSupplement Number 042 Issue 1 dated September 15 or later approved revision | Restricted to: G-TAMA ASN 051A, G-TAMB ASN 052A, G-TAMC ASN054A, G-TAMD ASN 056A & G-TAME ASN 035A | 03.10.2006 |
| 10027284 | AVIATION TRADERS LIMITED - ATL | P-EASA.A.S.03927 | | | A320-232 | Airbus A320 132C/8J/36Y LOPA | Type Certification Basis in accordance with the TCDS No. A.064 plus:- CRI D-2317 - "Application of Heat Release and Smoke Density Requirements to Seat Materials". CRI H-01/Special Condition H-01 "Enhanced Airworthiness Programme forAeroplane Systems - ICA on EWIS" | - Modification Summary Sheet, ref. doc. MOD-11490-00001, Issue 2, dated17th September 2009- Compliance Record, ref. doc. No. CR-11490-00001, Issue 2, dated 16thSeptember 2009 | The STC holder shall provide by 7th June 2010 Instructions for ContinuedAirworthiness on Electrical Wiring Interconnection System - EWIS ICA -in accordance with Special Condition H-01 "Enhanced AirworthinessProgramme for Aeroplane Systems - ICA on EWIS". | 18.09.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|-------------------|----------------|----------------------|---|--|--|--|---|---------------|
| 10027820 | AVIATION TRADERS LIMITED - ATL | P-EASA.A.S.03974 | | | 757-200 | 205 Passenger LOPA | As per TCDS | -Master Document List, MDL-11532-00001, Issue 1, 23/10/2009-Modification Summary Sheet, MOD-11532-0001, Issue 1, 23/10/2009Or later EASA approved revisions | The STC holder shall submit to the agency before 7th June 2010 final instructions of instructions for continued airworthiness on Electrical Wiring Interconnection System (E-WIS) in accordance with the special condition (H-01). | 08.11.2009 |
| 10027954 | AVIATION TRADERS LIMITED - ATL | P-EASA.A.S.03798 | REV. 1 | | BAE 146-200 | VIP Cabin Interior. Approval of Alternative Limitation on EASA STC10027954.EASA STC 10027954 contains the following Limitation: The STC holders shall submit to the agency before 7th June 2010 final instructions for continued airworthiness on Electrical (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. EASA STC 10027954 Special Condition H-01 Issue 1 Enhanced Airworthiness Programme for (...) | Modification Summary Sheet MOD-11193-00001 Issue 1 Master Document List MDL-11193-00001 or later revisions of the above listed documents approved by EASA | Serial Number E2079 only | 09.07.2010 |
| 10028747 | AVIATION TRADERS LIMITED - ATL | | REV. 1 | | 737 ALL MODELS | Blue Sky Network Satcom installation. This STC installs Satcom communications at the captain and F/O station. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-25 Amendment 7 The certified noise and/or emissions levels of the original product (...) | Modification Summary Sheet, ref : MOD-11619-00001 Issue 2 - Master Drawing List, ref : MDL-11619-00001 Issue 2- Aircraft Flight Manual Supplement, Ref : 11619-AFM-00001 Issue 1 or later revisions of the above listed documents approved by EASA | Not Applicable | 01.03.2010 |
| 10016672 | AVIDYNE CORPORATION | EASA.IM.A.S.03174 | | FAA STC SA00213CH | CIRRUS SR20 CIRRUS SR22 | Installation of Avidyne Model 9900BX/ TAS 600 Series Traffic Advisory System (TAS) | | According to Avidyne Master Document List Doc. No. DC1029-1 Rev. 12, or later FAA approved Revisions To be operated in accordance with Avidyne AFM-S Documents No. 32-2353C, Rev. 6 or later FAA approved Revisions To be maintained i.a.w. ICA Document No. (...) | 1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2) Th (...) | 27.05.2009 |
| 10030294 | AVIDYNE CORPORATION | | REV. 1 | FAA STC SA00269BO | 1. SR20, SR22 2. SR22T | Installation of Avidyne 700-00033(-) Tactical Weather Detection System | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Master Document List AVTWX-003, Rev. 14, Avidyne 700-00033(-) Tactical Weather Detection System-Instructions for Continued Airworthiness AVTWX-004, Rev. 7, Avidyne 700-00033(-) TWX-Avidyne Installation Manual 600-00165-000, Rev. 15 or later revisions (...) | As indicated in FAA STC SA00269BO | 21.12.2010 |
| 10033140 | AVIDYNE CORPORATION | | | FAA STC SA00296BO | SR20, SR22 | Installation of Avidyne Model DFC90 Autopilot P/N 700-00170-000 System | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | In accordance with Avidyne Master Document List AVDFC-019, Rev. 2 Avidyne AFM-S Document 600-00249-000, Rev. 01 Avidyne Instructions for Continued Airworthiness AVDFC-030, Rev. 00 Avidyne Installation Manual 600-00251-000, Rev. 02 Avidyne software P/N 53 (...) | as indicated in FAA STC SA00296BO Avidyne software P/N 530-00214(-) Rev. 02 or later FAA approved revision with underspeed protection functionality must be installed on PFD P/N 700-00006(-) | 20.12.2010 |
| 10038669 | AVIDYNE CORPORATION | | | FAA STC SA00310BO | PA46-350P, PA-46R-350T | Autopilot Series / DFC 90 Autopilot P/N 700-00170-002 DFC100 Autopilot P/N 700-00170-003 as replacement for the S-Tec System 55X Autopilot P/N1192(-) in accordance with Avidyne Corporation Master Document List AVDFC-077 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Note: The change has been classified and confirmed as "Basic" STC (non-significant). The (...) | Operation must be in accordance with Avidyne Corporation Aircraft Flight Manual Supplement (AFMS) as follows: For DFC90: Document 600-00285-000, Rev. 00, FAA approved on November 17, 2011 For DFC100: Document 600-00289-000, FAA approved on November 17, 2 (...) | Aircraft must be configured with an existing S-TEC 55X (or 55SR) Autopilot and with an existing Avidyne Flight Display System as follows in order to accomplish this STC: For DFC90: Avidyne Primary Flight Display, p/n 700-00006(-), loaded with Release 8. (...) | 06.03.2012 |
| 10039607 | AVIDYNE CORPORATION | | REV. 1 | FAA STC SA00282BO | PA-46R-350T | Integrated Flight Display | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. For those portions of the airplane that are unchanged from the basic PA-46R-350T the app (...) | Operation of the aircraft must be in accordance with Avidyne Airplane Flight Manual Supplement P/N 600-00263-001 Rev. 00, for Piper PA-46R-350T Matrix with Avidyne Integrated Flight Displays P/N 700-00063-XXX and Control Display Unit P/N 700-00160-XXX, (...) | Enhanced Vision System (EVS) is not included in the scope of this approval. Compatibility with Mode 5 alerts must be shown when a TAWS Class A is installed. | 11.10.2012 |
| 10041185 | AVIDYNE CORPORATION | | | FAA STC SA00282BO | SR20, SR22 | Installation of Avidyne Corporation 700-00083-XXX Entegra Integrated Flight Display System on Cirrus SR20 and SR22 according Avidyne Doc. No AVIFD-006 in latest FAA approved Revision | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. For those portions of the airplane that are unchanged from the basic SR20/ SR22 the app (...) | In accordance with Avidyne Master Document List AVIFD-005 Avidyne AFM-S Document 600-00712-000 Avidyne Instructions for Continued Airworthiness AVIFD-016 Avidyne Installation Manual 600-00160-000 or later revisions of the above listed documents approved (...) | As indicated in FAA STC SA00282BO Enhanced Vision System (EVS) is not included in the scope of this approval. Compatibility with Mode 5 alerts must be shown when a TAWS Class A is installed. | 28.08.2012 |
| 10042507 | AVIDYNE CORPORATION | | | FAA STC SA00323BO | 182E, 182F, 182G, 182H, 182J, 182K, 182L, 182M, 182N, 182P, 182R, 182S, 182T, R182. | Installation of Avidyne Corporation Autopilot; model DFC90, p/n 700-00170-022 (28-Volt Electrical system) or p/n 700-00170-024 (14-Volt Electrical system), loaded with Release 0 Software, as a replacement for the S-Tec Autopilot. | The Certification Basis of the original product and the following additional / alternative airworthiness requirements are applicable to this change: As specified in the original FAA STC No. SA00323BO dated 01-October-2012. The certified noise and/ or (...) | FAA approved S-TEC Master Document List No. AVDFC-115 rev.00, dated 25-Sep-2012. FAA Approved Airplane Flight Manual Supplement No. 600-00293-000, rev. 00 dated 01-October-2012. Instruction for Continued Airworthiness No AVDFC-125, rev.00 dated 09-Aug- (...) | Operation must be in accordance with Avidyne Corporation AFMS document No. 600-00293-000, rev.00 dated 01-October-2012. The AFMS must be carried in the Aircraft during all flights. Maintenance must be in accordance with ICA document AVDFC-125, rev.00da (...) | 05.12.2012 |
| 10042851 | AVION RESEARCH, Inc. | | REV. 1 | FAA STC SA170CH | 1. 172 1. 172A/B/C/D/E/F/G/H/I/K/L/M 2. 175, 175A/B/C | Installation of a Control Yoke and Instrument Panel in accordance with FAA STC SA170CH | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AVION Research International Drawing List C172ML0001, Rev. 2, dated May 5, 1994, FAA approved July 8, 1994 AVION Research International Installation Instructions C172/175-8001, Rev. 2, dated May 5, 1994, FAA approved July 8, 1994 or later revisions of th (...) | None | 21.12.2012 |
| 10014692 | AVIONES, PIEZAS Y ACCESORIOS, S.L. | EASA.A.S.02857 | | | PIPER PA-31 | Installation Dual integrated NAV/COMM/GPS GARMIN 480 and Transponder GARMIN GTX 33 (APA OT 5001 Rev.1). | | Supplement to Aircraft Flight Manual APA Document: SFM 5001 Rev. 0 Supplement Maintenance & Wiring Manual APA Document: SM&WM 5001 Rev. 0 | | 05.02.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|-------------------|----------------|----------------------|--|--|---|--|--|---------------|
| 10014704 | AVIONES, PIEZAS Y ACCESORIOS, S.L | EASA.A.S.02875 | | | MITSUBISHI MU-2B-35 | Installation of New Avionics Equipment to comply with COM/NAV regulations for IFR and B-RNAV Operation. | | New Supplement to Aircraft Flight Manual APA Document: SFM 6006 Rev. 2 (September 2006). Installation and Service Instructions:APA Document : OT 6006 Rev. 2 (September 2006). | | 10.12.2007 |
| 10017246 | AVIONES, PIEZAS Y ACCESORIOS, S.L | EASA.R.S.01443 | | | EUROCOPTER FRANCE AS 355 SERIE | HD I Ponti Radio Digital video, audio & data Transmission System. | | 0108002 revision 0 or later approved revision # Master Document List;0108002 revision 0 or later approved revision # Design Configuration;SFM 0108002 Revision 0 or later approved revision # Flight Manual Supplement;ICA 0108002 Revision 0 or later approved revision # (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 08.09.2008 |
| 10038296 | AVIONES, PIEZAS Y ACCESORIOS, S.L | DGAC-E 138 | REV. 1 | | 1. AB206 B 2. 206B, 206L | Transfer EASA STC 10038296 - Transfer from INAER MAINTENANCE S.A.U. to AVIONES, PIEZAS Y ACCESORIOS Power lines cleaning system with high pressure water with form "DGAC-E Certificate Data Sheet No.138" SLAAP is composed by a reservoir tank, a high pressu (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement (Bell 206-B series & Augusta Westland AB206-B series) INAER.M12/040/25-FMS-1 R3; Flight Manual Supplement (Bell 260L-3 & L-4) INAER.M12/040/25-FMS-2 R2; or later revisions of the above listed documents approved by EASA. Instructio (...) | None | 06.05.2013 |
| 10045232 | AVIONICARE LTD | | | | 727-100 | Upgrade of Dual Trimble HT-9100 GNSS to Dual Universal UNS-1Lw FMS on Boeing 727 Aircraft. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:- Special Condition(s): CRI H-01 'Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS' - the following par (...) | Avionicare Master Document List AVC-MDL-0004 Issue FAvionicare Flight Manual Supplement AVC-FMS-0004-001 Issue Bor later revisions of the above listed documents approved by EASA. | Applicable to s/n 18934 only Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 10.06.2013 |
| 10045968 | AVIONICS DESIGN SERVICES LTD. | | | TCCA STC SA 12-56 | DHC-8-102/-103/-106/-201/-202 DHC-8-301/-311/-314/-315 DHC-8-401/-402 | Apple iPad EFB Class 2 Provisions | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation must be in accordance with TCCA approved Avionics Designs Services Ltd. Master Drawing List MDL 11077, Revision C. Operation must be in accordance with TCCA approved Avionics Designs Services Ltd. Flight Manual Supplement FMS 11077-1, Revi (...) | Suitability of provisions has been demonstrated for Apple iPad 1, 2 and 3. This STC does not constitute an Operational Approval. Individual Operational Approval must be issued by the appropriate NAA. Prior to installation of this design change it must be (...) | 01.08.2013 |
| 10015426 | AVIONICS GROUP INC. | EASA.IM.A.S.01159 | REV. 1 | FAA STC ST02198AT | B757-200, -200CB, -200PF | Installation of a Teledyne Digital Flight Data Acquisition Unit, Part Number 2233000-811 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate as per the TCDS plus CRI H-01: ICA on EWIS. The requirements for environmental protection and the as (...) | Avionic Master Drawing List AG-31-1058.MDL, Revision B, 10/07/2007 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 08.02.2011 |
| 10016034 | AVIONICS GROUP INC. | EASA.IM.A.S.02203 | | ST02997AT | BOEING 737-100 BOEING 737-200 BOEING 737-300 BOEING 737-400 BOEING 737-500 | Installation of a Sandel Avionics ST3400 Class A Terrain Awareness and Warning System | | Avionics Group, Inc. Master Drawing List AG-34-1111.MDL, Revision IR, dated January 10, 2004, or later approved revision | FAA approved Airplane Flight Manual Supplement for Avionics Group, Inc. Sandel Avionics ST3400 Class A Terrain Awareness and Warning System dated December 8, 2004 is a required part of this STC. | 31.10.2006 |
| 10038339 | AVIONICS GROUP INC. | | | FAA STC ST03559AT | 767-200, -300, -300F, -400ER | Installation of Honeywell (TRA-67A) or Collins (TPR901) Mode S Enhanced Surveillance System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate. Certification Basis as per the TCDS and CS 25 Amendment 5 for EWIS requirements. The requirements to (...) | Avionics Group MDL AG-34-1223.MDL, Revision B, 12/03/2008 AFMS, AG-34-1233.AFM, Revision IR, 26/03/2008 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 08.02.2012 |
| 10014572 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.02676 | | | BAE 125-800 HAWKER 800 SERIES | Replacement of Dual FMS UNS-1D by Dual UNS-1F, to allow the aircraft to comply with PR-NAV requirements. | | Installation to be in accordance with AIEC Engineering Order ref AI569-062, rev A dated 05.09.2006 or later approved revision. Operation to be in accordance with AIEC Aircraft Flight Manual Supplement ref AI569-071, rev IR dated 14.06.2006 or later a (...) | | 21.09.2006 |
| 10014701 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.02870 | | | BEECH B200 SUPER KING AIR | Installation of a UAOS Flight Management System (FMS) UNS-1L | | Installation to be in accordance with AIEC Engineering Order ref CV631-062, rev. B dated 19/01/2007 or later approved revision. Operation to be in accordance with AIEC Aircraft Flight Manual Supplement ref CV631-071, rev IR dated 22/01/2007 or later a (...) | | 24.01.2007 |
| 10014739 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.02922 | | | BEECH B300 | Standby Instruments Installation: 1. 2" Revue Thommen Standby Altimeter & 2" Revue Thommen Airspeed Indicator & L3 Communications Standby Attitude Indicator AI-804 with Goodrich JET Emergency Power Supply Battery Pack PS-835 | | CV 596-049 | | 11.05.2007 |
| 10014793 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.02995 | REV 1 | | BEECH 400A | Upgrade of existing FID capable dual TDR-94D Mode S transponder system to enhanced surveillance capability by incorporation of Collins SB 502 and SB 507. | | # Master Data List, Doc. No. CV649-049, Revision IR, dated 19 January 2007 or later approved revision. # Airplane Flight Manual Supplement, Doc. No. CV649-071, Revision IR, dated 16 January 2007 or later approved revision. # Engineering Order (...) | 1. This change is limited to the configuration of aircraft / equipment type as listed under Item 1.2 in Engineering Order, Doc. No. CV649-062, Revision A, dated 23 March 2007, or later approved issue. 2. This STC is approved only for the produc (...) | 04.04.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|----------------|----------------|----------------------|---|--|---------------------|---|--|---------------|
| 10014815 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03023 | | | BOMBARDIER CL-600-2B16 | Enhanced Surveillance Mode S Transponder Upgrade (Collins) | | AEIC Modification AC652-015.AEIC MDL No. AC652-049 at Revision A dated March 7, 2007.AEIC AFM Supplement No. AC652-071 at Revision IR. | N/A | 15.03.2007 |
| 10014912 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03156 | REV 1 | | CESSNA 550 CITATION II | Honeywell ART 2000 Weather Radar transceiver/antenna installation interfacing with KMD 850 MFD to replace the existingRCA P-400 radar, to improve reliability and to add display of terrain and traffic graphics. | | Master Data list 679-049,AFM Supplement 679-071,ICA 679-085 ICA. | | 22.01.2008 |
| 10014913 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03157 | REV 1 | | CESSNA 550 CITATION II | Honeywell EGPWS Mark VIII System Installation,interfacing with KMD 850 MFD, as a replacement of the existing GPWS MK VI. | | Master Data list 681-049 Rev. D,AFM Supplement 681-071,ICA 681-085 ICA, Rev. A. | | 22.01.2008 |
| 10014917 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03163 | | | BEECH 200 | Upgrade of dual Mode S transponder system TDR-94 to TDR-94Dwith ELS capability and diversity | | Master Data List, Doc. No. 677-049, Rev. A, dated 09-Aug-2007, or later approved revisionEASA approved Airplane Flight Manual Supplement, Doc. No. 677-071, Initial Revision, dated15-Jun-2007, or later approved revisionInstructions for Continued Airw (...) | This change is limited to the configuration of aircraft / equipment type as listed under #1.2 Prerequisite Conditions#, in Engineering Order, Doc. No. 677-062, Initial Revision, dated 12-Jun-2007 | 14.08.2007 |
| 10014925 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03173 | | | BOMBARDIER CL-600-2B16 | Replacement of 3 FMS GPS Antennas (UASC) | | AEIC Modification Reference AC684-015,AEIC Master Data List, Document Number AC684-049 Revision IR dated 6 July 2007 or later approved revision,AEIC Instructions for Continued Airworthiness Document Number AC684-085 Revision IR dated 6 July 2007 or later approved revision. | | 04.09.2007 |
| 10014932 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03187 | | | BAE 125 -800A SERIES BAE 125-800B SERIES | Upgrade of EGPWS MK V to RAAS capability. | | EASA approved Master Data List, Doc. No. 687-049, Revision B, dated 18-Dec-2007, or later approved revisionEASA approved Airplane Flight Manual Supplement, Doc. No. 687-071, Revision A, dated 22-Oct-2007, or later approved revisionInstructions for C (...) | This STC is limited to the configuration of aircraft / equipment type as listed under item 2.1 Limitation, in Engineering Order, Doc. No. 687- 062, Revision B, dated 18-Dec-2007, or later approved revision. | 10.01.2008 |
| 10014959 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03230 | | | BAE 125 -800 SERIES | Modification of Collins Transponders to achieve Enhanced Surveillance Capability. | | Installation in accordance with AEIC Engineering Order 666-062 latest approved revision. Aircraft to be operated in accordance AEIC Aircraft Flight Manual Supplement, Document ref 666-071 latest approved revision,Aircraft to be Maintained in accordance (...) | None. | 10.09.2007 |
| 10015026 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03333 | | | BOMBARDIER CL-600-2B19 | Dual Electronic Flight Bag Installation (Maxware). | | Master Data List 667-049. | Aircraft Flight Manual Supplement 667-071. | 11.04.2008 |
| 10015030 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03342 | | FOCA STC 34-45-004 | CESSNA 525, CITATION JET CESSNA 525A | TCAS-400 & dual TDR-94D installation. Change to the FOCA STC34-45-004 TCAS-4000, Dual TDR-94D, GNS-430 Installation. By incorporation of an Gables IST.G7614-() as an alternative ATC/TCAS controller. | | Engineering Order, Doc. No. AF584-062.01 | | 10.10.2007 |
| 10015114 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03462 | | | LEARJET 60 | UNS-1B or 1C upgrade to UNS 1C+ | | Master Data List Doc. No.: 734-049 AFM Supplement Doc. No. 734-071 | | 25.04.2008 |
| 10015143 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03503 | | | 1329-25 | Upgraded Avionics systems:- UNS-1D to UNS-1D+ FMS- Dual VHF-20A system to VHF-22C incl CTL-22C Control Head- Dual DME-42 system with 2 CTL-32C Control Head- Dual VIR-30 NAV system with 2 CAD-31 Control Adapter- Altimetry system with a IS&SAir Data Sy (...) | | - Master Data List, Doc. No. AD741-049, Rev. IR or latest approved revision- Master Data List, Doc. No. AD744-049, Rev. IR or latest approved revision- AFMS, Doc. No. AD744-071, Rev. IR or latest approved revision- AFMS, Doc. No. 03-3510-01/05, Rev. (...) | Reg.: HB-JGK S/N: 5233 valid for this S/N only. | 14.07.2008 |
| 10015155 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03520 | | | LEARJET 60 | Installation of Iridium SAT/COM Antenna (Sensor Systems) | | As noted in Master Data List 745-049, rev. IR (or later approved revisions) | Installation of the Iridium SAT/COM antenna and the antennacoax outlet as covered by this STC does not constitute an approval of the utility system installation or operation. | 22.04.2008 |
| 10015213 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03628 | | | LEARJET 31 | P-RNAV Compliance Demonstration | | As per Master Document List EE759-049, rev. IR (or later approved revisions) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 25.08.2008 |
| 10015252 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03697 | | | BOMBARDIER CL-600-2B16 | P-RNAV Compliance | | Master Data List No. AC780-049,AFM Supplement Change Sheet No. AC780-071 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 13.10.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|------------------|----------------|----------------------|-------------------|---|--|---|---|---------------|
| 10015260 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03718 | REV 1 | | BEECH 390 | Indium SATCOM System Installation (Sky Connect) | | Master Data List, Doc. No. 788-049, Revision B, dated 09-Feb-2009, or later approved revision/EASA approved Pilots Information Sheet, Doc. No. 788-091, Revision A, dated 09-Feb-2009, or later approved revision/Instructions for Continued Airworthiness. (...) | This change is limited to the configuration of aircraft / equipment type as listed under 3.1 Mechanical Installations# in Engineering Order, Doc. No. 788-062, Revision A, dated 13- Jan-2009, or later approved revision.1 Prior to installation (...) | 09.02.2009 |
| 10015280 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03806 | REV. 2 | | CESSNA 206, 210 | Installation of Aspen Avionics EFD 1000 system (STC to cover now CessnaC-206 & newly added Cessna C-210 series) | FAR 23The Certification Basis for the original product remains applicable to this certificate/ approval.Minor Change to EASA.A.S.03806 Revision 1 to include C-210 seriesThe certified noise and/ or emissions levels of the original product are unchanged (...) | Master Data List: Doc Number 801-049 Rev. BAFMS EFD 1000 System: Doc Number 801-071 Rev. B.Instructions for continued airworthiness: Doc Number 801-085 Rev. B | Not valid if the Garmin 1000 ICS option is installed. | 25.06.2010 |
| 10015292 | AVIONICS INTEGRATION & ENGINEERING | EASA.A.S.03834 | | | CESSNA 206 SERIES | Dual GNS 430w with TAS 600 SystemsBy installing two GARMIN GNS430W with external 121.5MHz Frequency selection and an Avidyne TAS 600 system | | Master Data List: Doc Number 808-049AFMS Dual GNS430W with TAS600: Doc Number 808-071Instruction for continued Airworthiness: Doc Number 808-085 | Aspen EFD-1000 Primary Flight Display must be installed in the aircraft. | 20.04.2009 |
| 10017244 | AVIONICS INTEGRATION & ENGINEERING | EASA.R.S.01441 | | | A 109 K2 | Replacement Euronav III by Euronav V (Euroavionics)By removing- the existing RN 5 Euronav III Map Generatorand installing- a new RN 6 Euronav V Digital Map Generator | | - Master Data List, Doc. No. 725-049, Rev. IR, or latest approved revision- RFMS, Doc. No. 725-071, Rev. IR, or latest approved revision | Prerequisite conditions:- Dual GNS-530A installation- RN 5 Euronav III Map Generator installed with interface to the GNS-530A 2- CLN.14-3.0 Control Unit installed- DYN764-2.0 Display installed | 28.04.2008 |
| 10017268 | AVIONICS INTEGRATION & ENGINEERING | EASA.R.S.01488 | | | EC-135-T2 | COMNAV/ATC Avionics System Installation. | | AIEC Master Data List ED743-049, Rev. B or later approved Revision. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 04.02.2009 |
| 10028083 | AVIONICS INTEGRATION & ENGINEERING | P-EASA.A.S.03930 | | | LEARJET 35A | Installation of Artex ELT C406-02 (Cobham). | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approval.As specified in Certification Program 900-080, rev. A (or later approvedrevisions).The (...) | As specified in MDL 900-049, rev. Aor later revisions of the above listed documents approved by EASA | None. | 01.12.2009 |
| 10028084 | AVIONICS INTEGRATION & ENGINEERING | P-EASA.A.S.03936 | | | LEARJET 35A | Installation of ART 2000 Weather Radar with KMD 850 MFD (Honeywell). | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approval.As specified in Certification Program 912-080, rev. A (or later approvedrevisions).The (...) | As specified in MDL 912-049, rev. Aor later revisions of the above listed documents approved by EASA | None. | 01.12.2009 |
| 10028085 | AVIONICS INTEGRATION & ENGINEERING | P-EASA.A.S.03933 | | | LEARJET 35A | Installation of SSFDR F1000 (L-3 Communications). | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approval.As specified in Certification Program 846-080, rev. A (or later approvedrevisions).The (...) | As specified in MDL 848-049, rev. Aor later revisions of the above listed documents approved by EASA | None. | 01.12.2009 |
| 10028086 | AVIONICS INTEGRATION & ENGINEERING | P-EASA.A.S.03932 | | | LEARJET 35A | Installation of EGPWS Mk VIII (Honeywell). | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approval.As specified in Certification Program 911-080, rev. A (or later approvedrevisions).The (...) | As specified in MDL 911-049, rev. Aor later revisions of the above listed documents approved by EASA | None. | 01.12.2009 |
| 10030099 | AVIONICS INTEGRATION & ENGINEERING | | | | MBB-BK 117 C-1 | Cabin Interior RetrofitInstallation of a cabin interior retrofit based on AIEC engineeringorder 854-062. | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval | MBB-BK117 C-1 Rotorcraft Flight Manual Supplement - Doc. Number: 854-071Rev.: Aor later revisions of the above listed documents approved by EASACertification Compliance Sheet - Doc. 854-075 Rev. B, 19.05.2010Certification Program - Doc. 854-080 Rev. (...) | S/N 7541 | 21.05.2010 |
| 10030961 | AVIONICS INTEGRATION & ENGINEERING | | | | BEECH 200 | Installation of HySpex Camera System (NEO) | The Certification Basis of the original product remains applicable to this change.The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -EASA approved Master Data List, Doc. No. CV877-049, Rev. A, dated14-Jul-2010, or later approved revision-EASA approvedEngineering Order, Doc. no. CV877-062, Rev. A, dated14-Jul-2010, or later approved revision-EASA approved Airplane Flight Manual S (...) | Only aeroplanes in compliance with the preconditions as listed inChapter C of Engineering Order, Doc. No. CV877-062, Revision A, dated14-Jul-2010, or later approved revision, are eligible for this change. | 16.07.2010 |
| 10032056 | AVIONICS INTEGRATION & ENGINEERING | | | | 35A / 36A | COMJAM System Installation | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approval.Refer to Certification Program 856-080, rev. IRThe requirements for environmental prot (...) | Refer to Master Data List 856-049(>,<,> Rev. A or later revisions of the above listed documents approved by EASA | This COMJAM system installation approval is only applicable to the following aircraft equipped with certain special mission provisions (including electrical power supply, harnesses and wiring, workstationand antennas): s/n's 35A-139, 35A-179, 35A-268, 35A-331, 35A-387,36A-062 and 36A-063. | 05.10.2010 |
| 10033418 | AVIONICS INTEGRATION & ENGINEERING | P-EASA.A.S.03870 | | | PA-31T | TAS 620 System Installation by installing an Avidyne TAS 620 TrafficAdvisory System | FAR 23 | - Master Data List, Doc. No. 801-049, Rev. A or latest approvedrevision- AFMS, Doc. No. 818-071, Rev. A or latest approved revision- ICA, Doc. No. 818-085, Rev. IR or latest approved revision | None | 03.07.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|-----------------|----------------|----------------------|------------------|---|---|--|--|---------------|
| 1003543 | AVIONICS INTEGRATION & ENGINEERING | | | | EC135 P2+ | GNS-530 & GNS-430A Installation/Out: Garmin GNS-530W & GNS-430W/In: Garmin GNS-530 & GNS-430A Complete retrofit done by American Eurocopter under FAA Form 337 dated 11.12.2009 and FAA Form 8110-3 dated 11.12.2009 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List Doc. No.: EV5126-049 Rev. IR Instructions for Continued Airworthiness Doc. No.: EV5126-085 Rev. IR Rotorcraft Flight Manual Supplement Doc. No.: EV5126-071 01 Rev. IR or later revisions of the above listed documents approved by EASA | This EASA STC is only valid after EASA Validation of FAA STC SR09137RC-D (Extended R/H Side Instrument Panel) S/N: 0664 | 29.06.2011 |
| 10037002 | AVIONICS INTEGRATION & ENGINEERING | | | | HAWKER 900XP | Cabin Interior Removal and Modification. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Refer to Certification Program 5143-080, rev. B The requirements for environmental protection (...) | Modification in accordance with Master Data List Doc. No. 5143-049, Rev. A Instructions for Continued Airworthiness ref Doc. No. 5143-085, Rev. IR or later revisions of the above listed documents approved by EASA. | None | 24.10.2011 |
| 10038857 | AVIONICS INTEGRATION & ENGINEERING | | | | CL-600-2B16 | Indium Satcom Installation (ICG) | EASA IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged (...) | AIEC Corp Master Data List 5163-049 revision B dated 20/03/2012 AIEC Corp Engineering Order 5163-062 revision A dated 27/01/2012 AIEC Corp Instructions for Continued Airworthiness 5163-085 revision A dated 27/01/2012 AIEC Corp AFM Supplement 5163-071 (...) | None | 20.03.2012 |
| 10038859 | AVIONICS INTEGRATION & ENGINEERING | | | | CL-600-2B16 | Terrain Awareness and Warning System (TAWS) Upgrade to SCN 12.0 (UASC) | EASA IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged (...) | AIEC Corporation Master Data List No. 5173-049, Revision B, dated 19 March 2012 AIEC Corporation Engineering Order No. 5173-062, Revision A, dated 16 January 2012 AIEC Corporation Instructions for Continued Airworthiness No. 5173-085, Revision A, dated (...) | None | 20.03.2012 |
| 10038860 | AVIONICS INTEGRATION & ENGINEERING | | | | CL-600-2B16 | CVR-120 replacement by CVR-120R (UASC) | EASA IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged (...) | AIEC Corporation Master Data List No. AC5136-049, Revision B, dated 14 March 2012 AIEC Corporation Engineering Order No. AC5136-062, Revision IR, dated 16 June 2011 AIEC Corporation Instructions for Continued Airworthiness No. AC5136-085, Revision IR, d (...) | None | 20.03.2012 |
| 10039495 | AVIONICS INTEGRATION & ENGINEERING | | REV. 1 | | 35A/36A | Enhanced Surveillance Upgrade to MST 67A Transponder (Honeywell) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AIEC Master Data List: ref 5187-049, rev A, date 17 April 2012 AIEC AFM Supplement: ref 5187-071, rev IR, date 26 March 2012 AIEC Instructions for Continued Airworthiness: ref 5187-085, rev IR, date 23 March 2012; or later revisions of the above listed (...) | None | 16.05.2012 |
| 10039496 | AVIONICS INTEGRATION & ENGINEERING | | | | 35A/36A | Replacement of UNS-1M by UNS-1Fw MMMS (UASC) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AIEC Master Drawing List: ref 5183-049, rev A, dated 26 April 2012 AIEC AFM Supplement: ref 5183-071, rev B, dated 19 April 2012 AIEC AFM Supplement: ref 5183-071 01 rev IR, dated 11 April 2012 AIEC Instructions for Continued Airworthiness: ref 5183- (...) | None | 08.05.2012 |
| 10041282 | AVIONICS INTEGRATION & ENGINEERING | | | | CL-600-2B16 | CPDLC System Capability | FAR 25 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and (...) | Master Data List: AC5211-049 Rev B dated 03 Sep 12 Airplane Flight Manual Supplement: AC5211-071 Rev B Instructions for Continued Airworthiness: AC5211-085 Rev A dated 27 Aug 12 Compliance Declaration: AC5211-016 Rev IR dated 03 Sep 12 Certification Compliance (...) | CPDLC system not compliant with Commission Regulation (EC) 29/2009 | 04.09.2012 |
| 10042908 | AVIONICS INTEGRATION & ENGINEERING | | REV. 1 | | L35A, L36A | Multiple Antenna Installation Provision Lear 35A/ 36A Antennas are installed at the following location: 1. UHF antennas at the bottom fuselage between FR 13 and 13E-As option A: antenna located between FR 13B and 13A center fuselage.-As option B: ante (...) | FAR 25 The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ICA Instruction for continued airworthiness Document No. 5223-085 revision IR or later revisions of the above listed documents approved by EASA | None | 07.01.2013 |
| 10043036 | AVIONICS INTEGRATION & ENGINEERING | P-EASAR.S.01381 | | | MBB-BK 117 B-2 | Avionics Retrofit Out: ADF System King KDF-806; Dual VHF-COM System Becker AR-3202; VHF-NAV System King KNS-81; VHF-NAV System Becker NR-3300-2. In: Dual VHF-COM/NAV System King KX-165; Marker System King KR-21; ADF System King KR-87; Transponder Mode (...) | | Master Data List MDL Doc. No. 662-049- Rotorcraft Flight Manual Supplement Doc. 662-071 Rev. IR or later approved revisions. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The ins (...) | 12.11.2007 |
| 10044866 | AVIONICS INTEGRATION & ENGINEERING | | | | LEARJET MODEL 45 | ISAT Satcom Antenna Installation (Reference Master data list Doc. No. 5240-049) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: JAR 25 Change 13 CR1 special condition A10 The requirements for environmental protection (...) | ICA Instruction for Continued Airworthiness Doc. 5240-085 ALS mandatory structural inspection requirement Master data list Doc. No. 5240-049 or later revisions of the above listed documents approved by EASA. | None | 14.05.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|---------------------|----------------|----------------------|---|---|---|---|---|---------------|
| 10045086 | AVIONICS INTEGRATION & ENGINEERING | | | | LEARJET 31A/35A/36A | SM GPS Antenna ProvisionReference Master data list Doc. No.5256-049 | FAR 25 Amdt. 23The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - ICA Instruction for Continued Airworthiness Doc. 5256-085- Master data list Doc. No. 5256-049or later revisions of the above listed documents approved by EASA | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 29.05.2013 |
| 10015706 | AVIONICS SUPPORT GROUP, INC | EASA.IM.A.S.01696 | | FAA STC ST03151AT | B737 SERIES | Installation of an Avionics, Inc. Secure LINK System for ground only operation | | Avionics Support Group, Inc. Master Data List 25817-MDL, Rev. IR, dated 15th June 2005, or later approved revisions | | 17.01.2007 |
| 10015345 | AVIONICS SUPPORT GROUP, INC. | EASA.IM.A.S.00446 | | FAA STC ST02900AT | BOEING 747-200 BOEING 747-300 | T2CAS provisions on Boeing 747-200 & -300 Series Aircraft. | | - FAA STC ST02900AT - Avionics Support group Inc Master Drawing List 24595-MDL, Revision IR dated June 22, 2004 or later approved revision. | 1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated shall remain valid until surrendered, withdrawn or otherwise terminated.2. If transfer of this STC is requested under Part 21A.116, the STC will be reissued.3 (...) | 21.12.2004 |
| 10015447 | AVIONICS SUPPORT GROUP, INC. | EASA.IM.A.S.01188 | | FAA STC ST02976AT | BOEING 737-300 SERIES BOEING 737-400 SERIES BOEING 737-500 SERIES | Installation of ACCS Elementary and Enhanced Surveillance capable Mode S Transponders and Flight ID capable Transponder/Control Panel | | FAA STC ST02976AT dated 10th November 2004ASG Master Drawing List, Ref 24640, Rev A dated 4th February 2005ASG AirplaneFlight Manual Supplement, Ref 24640-FMS, Rev IR dated 10th November 2004ASG Instruction for Continued Airworthiness, Ref 24640-CAW, Rev IR dated 27th September 2004 | 1. According to FAA STC No ST02976AT2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.10.2005 |
| 10015660 | AVIONICS SUPPORT GROUP, INC. | EASA.IM.A.S.01637 | | FAA STC ST02763AT | BOEING 757-200 BOEING 757-300 | Installation of an In-Flight Cabin Video Surveillance System | | - Avionics Support Group Inc Master Drawing List 23485-MDL, Revision IR dated October 11, 2003 or later approved revision.- Avionics Support Group Inc AFM Supplement reference 23485 Revision IR or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 16.03.2006 |
| 10016116 | AVIONICS SUPPORT GROUP, INC. | EASA.IM.A.S.02307 | | FAA STC ST03071AT | BOEING 727-200 | Installation of Avionics System for Reduced Vertical Separation Minimum (RVSM) Provisions into a Boeing 727-200 aircraft. | | Avionics Support Group, Inc Master Drawing List (MDL) No. 23509-MDL, Rev A, dated 27/05/2004, or later approved revisions | | 04.06.2007 |
| 10026764 | AVIONICS SUPPORT GROUP, INC. | P-EASA.IM.A.S.03249 | | FAA STC ST03700AT | 747-400, 747-400F | Installation of an International communications Group ICS-220A IridiumSATCOM System in a B747-400/-400F series aircraft in accordance with Avionics Support Group, Inc. Master Data List 28546-MDL, Revision A, dated March 16, 2009, or later FAA approved revision.Validation of FAA STC03700AT | Per FAA TCDS A20WE | -28546-AFMS Rev. IR dated May 14, 2009 Airplane Flight Manual Supplement-ASG Inc Aircraft Maintenance Manual Supplement23-15-00-DER Services, Inc. Report SR-A-2008-083-2 Rev. IR Damage Tolerance Analyses for Iridium Antenna Installation. | After installation of this STC the operator is required to incorporate in the Airworthiness Limitations Section of the Instructions for Continued Airworthiness applicable to the existing Maintenance Program a detailed inspection item at a threshold of (...) | 10.08.2009 |
| 10013962 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.00646 | REV. 1 | | | Installation of single/dual Garmin GTX330D. | | 1. Installation in accordance with ASR-2007-079-IF1-10-00 and STC MDL ASR-2007-079-MDL-01-02, or later approved Revisions.2. To be operated in accordance with Avionik Straubing Airplane Flight Manual Supplement Doc-No. ASR-2007-079-FMS-05.1.00 (...) | | 17.06.2008 |
| 10014031 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.01097 | | | PIPER PA-31 | Installations of Avidyne FM850, 2x Garmin GTX330D Mode S Transponder and Garmin GMA340 Marker/Audio | | EASA approved AFM Supplements:Avidyne FM850 Flight Display LBA2004-024-01 Rev AGarmin GTX330D Mode S Transponder LBA2004-024-00 Rev AGarmin GMA340 Marker/Audio LBA2004-024-02 Rev A or later EASA approved issues | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 26.10.2005 |
| 10014032 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.01098 | | | BEECH 300LW | Installation of Mode S Transponder TDR94D with new Top Antenna. | | AFM Supplement Mode S Transponder Collins TDR 94 D' Doc. No.: ACG 2004-014-00Rev. A, Engineering Order ACG2004-014-00 | | 22.06.2005 |
| 10014089 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.01185 | | | CESSNA 501 CESSNA 551 | Installation of Artex ELT C406-1. | | EASA-approved Flight Manual Supplement FMS ASR 2005-007-00, Rev. A, or later EASA-approved revision. | | 28.04.2005 |
| 10014174 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.01309 | | | BEECH 85-90 BEECH 65-A90 BEECH 90 BEECH B90 BEECH C90 BEECH C90 A BEECH E90 | Installation of Skywatch L3 SKY899, Radar RDR 2000 and EGWPS KGP 560 | | EASA approved AFM Supplement FMS ASR 2005-020-00 Rev. A 'Skywatch 899', or later EASA approved revisionsEASA approved AFM Supplement FMS ASR 2005-020-01 Rev. A 'RDR2000', or laterEASA approved revisionsEASA approved AFM Supplement FMS ASR2005-020-02 Rev. A 'EGWPS Honeywell KGP 560', or later EASA | | 07.10.2005 |
| 10014213 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.01372 | | | CESSNA 501 CESSNA 551 | Installation of Dual Garmin GNS 430A | | EASA approved AFM Supplement EASA2005-032-00 Rev. A, or later EASA approved issues | | 01.09.2005 |
| 10014273 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.01507 | | | BEECH 90 | Installation of Avidyne FM850 | | -EASA approved AFM Supplement LBA2004-015-00 Rev A or laterEASA approved issues | | 18.08.2005 |
| 10014311 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.01556 | | | CESSNA 525 | Installation of SKY497 and TAWS 8000The SKY497 is wired to the new Antenna, existing AM-250 Copilot Altimeter, I-linc MFD and Audio CTL. The TAWS 8000 is wired to the existing MFD, KLN90B and ADC 200 | | EASA approved AFM Supplement FMS ASR 2005-023-00 Rev. A, or later approved revisions | | 21.06.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|----------------|----------------|----------------------|---|---|---------------------|--|--|---------------|
| 10014340 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.01598 | | | CESSNA 550 | Installation of CAS67A (incl. MST67A), Mark VIII, Karnaad 406AF and Garmin GPS400 | | Table of Contents FO E-05 (ASR 2005-042-00) Rev. 0 or later approved Revision Flight Manual Supplement FMS-ASR2005-042-00 Rev. A (TCAS II CAS67A) or later approved Revision Flight Manual Supplement FMS-ASR2005-042-01 Rev. A (XPDR MST67A) or later approv (...) | Certification Basis of the Change: CS 25, additionally AMC 20-4 (Basic R-Nav), AMC 20-5 (GPS), JAA Leaflet No. 8, Rev. 2 (ACAS II), JAA Leaflet No. 12 (TAWS), JAA Leaflet No. 13 Rev. 1 (Mode-S ELS), JAA Leaflet No. 18 (Mode-S EHS) | 11.04.2006 |
| 10014448 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.02501 | | | PIPER PA-46-310P | Installation of GNS530, with GDC31 Roll Steering Converter | | FMS ASR 2006-004-01 Rev. 1, Garmin GNS530/FMS ASR 2006-004-02 Rev. 1, DAC GDC31 | | 05.04.2006 |
| 10014546 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.02634 | | | PIPER PA-31P | Installation of COM/NAV/GS/GPS Garmin GNS530/GNS430 (dual installation) | | EASA Approved Flight Manual Supplement ASR 2006-025-00 Revision A or later EASA approved revision | | 11.09.2006 |
| 10014566 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.02667 | | | PIPER PA34-200 PIPER PA34-200T PIPER PA34-220T | Installation of dual Garmin GNS430 and Dittel FSG 2T in a Piper PA34. | | Engineering Order ASR 2006-029-00 Rev. A Master Drawing List AFM Supplement ASR 2006-029-01 Rev. A AFM Supplement ASR 2006-029-02 Rev. A ICA Garmin GNS 430 dual installation ICA Dittel FSG 2T installation ICA ACK A-30 Altitude Digitizer Installation Instructions ASR2006-029-00 10 | 1. Maximum operating altitude 18.000 ft2. GPS is not approved for TERM and APPR NAV3. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility w (...) | 22.05.2006 |
| 10014615 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.02744 | | | PIPER PA-46-310P PIPER PA-46-350P | Installation of COM/NAV/GS/GPS GNS430, with GDC31 Roll Steering Converter. Installation of Diversity Mode S XPDR Garmin GTX330D. Installation of TCAS Avidyne TAS610. | | # FMS ASR 2006-041-01, Rev. A Garmin GNS 430# FMS ASR2006-041-02, Rev. A DAC GDC31# FMS ASR 2006-041-03, Rev. A Garmin GTX 330D# FMS ASR 2006-041-04, Rev. A Avidyne TAS 610# FMS ASR 2006-041-05, Rev. A COM 2 GND Clearance Function | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 07.05.2007 |
| 10014705 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.02876 | | | BEECH 65-90 BEECH 65-A90 BEECH B90 BEECH C90 BEECH C90A BEECH E90 | Dual Installation of COM/NAV/GS/GPS GNS430, with GDC31 Roll Steering Converter. Installation of Stormscope WX500 and TCAS Avidyne TAS610. | | # Installation must be in accordance with Master Document List Engineering Order No. ASR 2006-067-00 Rev. A dated 07-07-07 or later Approved Revisions.# Operation must be in accordance with Flight Manual Supplement No. ASR 2006-067-01 to ASR 2006-067-02 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 12.07.2007 |
| 10014772 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.02964 | | | FAIRCHILD SWEARINGEN SA 226-AT FAIRCHILD SWEARINGEN SA 226-T FAIRCHILD SWEARINGEN SA 226-TI FAIRCHILD SWEARINGEN SA 227-AT | Installation of Mode S XPDR Diversity Garmin GTX 330D with now top Antenna. | | Doc. No. ASR 2006-073-MDL-01-00 Amendm. 00 dated 24/08/2007 or later approved Revisions. Operation must be in accordance with Avionik Straubing Flight Manual Supplement No. ASR 2006-073-FMS-05-00 Amendm. 00 dated 27/08/2007 or later approved versions. M (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 16.10.2007 |
| 10014773 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.02965 | | | CESSNA 525 | Installation of one Garmin GNS530 in a Cessna 525. | | Installation must be in accordance with Avionik Straubing Master Document List Doc. No. ASR 2008-021-MDL-01-00 Amendm. 00 dated 01.04.08 or later Approved Revisions. Operation must be in accordance with Avionik Straubing Flight Manual Supplement No. A (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 19.09.2008 |
| 10014802 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.03005 | | | CESSNA 525 CESSNA 525A | Installation of Garmin GTX330D single/dual. | | Installation must be in accordance with Master Document List Engineering Order No. ASR 2007-006-00 Rev. 00 dated 25-01-07 or later Approved Revisions. Operation must be in accordance with Flight Manual Supplement No. ASR 2007-006-FMS-05.1-00 Rev. 00 dated 25-01-07 or later approved revision. | Limited to Cessna 525, SN 0001-0359. Compatibility of this modification with other previously approved modifications must be determined by the installer. | 08.02.2007 |
| 10014811 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.03016 | | | CESSNA 525A | Installation of single/ dual Garmin GTX 330D Model: Cessna 525A, S/N 0001 and on | | ASR2007-009-TCP-06.00, ASR2007-009-MDL-01-01 EASA approved AFM-S Doc. No. ASR2007-009-FMS-05.1-00, Issue 1, or later approved Revision EASA approved AFM-S Doc. No. ASR2007-009-FMS-05.2-00, Issue 1, or later approved Revision EASA approved AFM-S Doc. No (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 21.06.2007 |
| 10015159 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.03528 | | | FAIRCHILD SWEARINGEN SA 226-AT FAIRCHILD SWEARINGEN SA 226-T FAIRCHILD SWEARINGEN SA 226-TI FAIRCHILD SWEARINGEN SA 226-TC FAIRCHILD SWEARINGEN SA 227-AC FAIRCHILD SWEARINGEN SA 227-BC FAIRCHILD SWEARINGEN SA 227-DC | Installation of honeywell Mark VI EG PWS - Optional installation of Comant CI401-220 GPS Antenna. | | Installation must be in accordance with Avionik Straubing Master Document List Doc. No. ASR 2007-144-MDL-01-00 Amendm. 00 dated 20.12.07 or later Approved Revisions. Operation must be in accordance with Avionik Straubing Flight Manual Supplement No. A (...) | | 06.05.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|------------------|----------------|----------------------|--|---|---|---|--|---------------|
| 10015229 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.03647 | | | CESSNA 501 CESSNA 551 | Installation of COMNAV/GS/GPS Garmin GNS430W (AW) or COMNAV/GS/GNS530W (AW). | | 1. According to Avionik Straubing Master Document List Doc. No. ASR-2008-065-MDL-01-00 Rev. 00, or later Revisions.2. To be operated in accordance with approved Avionik Straubing AFM-S Document No. ASR-2008-065-FMS-05-00, Rev. 00, or later a (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 18.11.2008 |
| 10015244 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | EASA.A.S.03679 | | | CESSNA 401 CESSNA 401A CESSNA 401B CESSNA 402 CESSNA 402A CESSNA 402B CESSNA 402C CESSNA 404 CESSNA 406 CESSNA 411 CESSNA 411A CESSNA 414 CESSNA 414A CESSNA 421 CESSNA 421A CESSNA 421B CESSNA 421C CESSNA 425 CESSNA 441 SERIES CESSNA F406 | Installation of Single or Dual Garmin G600 Avionics System and dual Garmin GNS4/530(A)W GPS/COMM/NAV | | -Project Description: ASR-2008-104-PRD-04-00, Rev. 00-Instructions for Installation: ASR-2008-104-IFI-10-00, Rev. 00-Instructions for Continued Airworthiness: ASR-2008-104-ICA-11-00, Rev. 00-Flight Manual Supplement: ASR-2008-104-FMS-05-00, Rev. 00 | 1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2) Th (...) | 15.06.2009 |
| 10026784 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | P-EASA.A.S.03917 | REV. 1 | | PIPER PA-46-310P PIPER PA-46-350P PIPER PA-46-500TP PIPER PA-46R-350T | Installation of Single or Dual Garmin G500 / G600 Avionics System(including SVS) and Dual Garmin GNS4/530(A)W GPS/COMM/NAV. | The Certification Basis of the original product and the followingadditional / alternative airworthiness requirements are applicable tothis change- For those areas affected by the change compliance showing against CS23has been elected.- EASA Guidance (...) | -EASA approved Master Document List, Doc. No. ASR-2009-075-MDL-01-00.Amdt. Level 00, dated 20-Jan-2010, or later approved amdt. level.EASA approved Instructions for Installation, Doc. No. ASR-2009-075-IFI-10-00, Amdt. Level 00, dated 18-Jun-2009, or (...) | This STC is limited to the configuration of aircraft, equipment type, and functions as listed in EASA approved Instructions for Installation, Doc. No. ASR-2009-075-IFI-10-00, Amdt. Level 00, dated 18-Jun-2009, especially as mentioned in Item 2. of this (...) | 23.04.2010 |
| 10026941 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | P-EASA.A.S.03675 | | | BEECH 60, A60, B60 | Installation of Garmin GNS430W / GNS430AW / GNS430 / GNS530W / GNS530AW/ GNS530, with Installation of Garmin GA35/GA56 GPS Antenna. | CS-23 | Installation must be in accordance with Master Document List Engineering Order No. ASR 2006-088-MDL-01-00 Amendment 00 dated 08-01-2009 or later Approved Revisions. Operation must be in accordance with Flight Manual Supplement No. ASR2007-088-FMS-05.1-0 (...) | None | 25.08.2009 |
| 10028197 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | P-EASA.A.S.03875 | | | B90, C90, C90A, C90GT, C90GT1, E90, F90 | Installation of an FMS System Universal UNS-1Lw and an EGPWS KGP 560 | CS-23 Amdt. 1The Certification Basis for the original product remains applicable tothis certificate/ approvalThe Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis (...) | -Instructions for Installation ASR-2009-031-IFI-10-00-AFM-Supplement ASR-2009-031-FMS-05.1-00 (UNS-1Lw)-AFM-Supplement ASR-2009-031-FMS-05.2-00 (KGP 560)- Instructions for Continued Airworthiness ASR-2009-031-ICM-11.1-00(UNS-1Lw)-Instructions for Con (...) | None | 08.12.2009 |
| 10028486 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | SEE ATTACHED MODEL LIST | Installation of an Aspen EFD-1000 ProSystem. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS-23, Initial Issue (not required by 21A.101, elected to comply by theapplicant).The (...) | Aircraft Model List: ASR-2009-047-AML-24-00 (attached)Installation Instructions: ASR-2009-047-IFI-10-00Instructions for Continued Airworthiness: ASR-2009-047-ICA-11-00Flight Manual Supplement: ASR-2009-047-FMS-05-00or later revisions of the above listed documents approved by EASA | None | 15.01.2010 |
| 10028816 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | P-EASA.A.S.03916 | | | MOONEY M20 C, E, F, G, J, K MOONEY M20 L, M, R, S | Installation of Aspen EFD1000 PRO System. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS-23, Initial Issue (not required by 21A.101, elected to comply by theapplicant)The c (...) | Instructions for Installation: ASR-2009-046-IFI-10-00, Rev. : 00Instructions for Continued Airworthiness: ASR-2009-046-ICA-11-00, Rev.00Flight Manual Supplement: ASR-2009-046-FMS-05-00, Rev. 00or later revisions of the above listed documents approved by EASA | None | 11.02.2010 |
| 10029814 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | | | | PA-46-310P, -350P, PA-46-500TP, PA-46R-350T | Installation of a Stormscope Avidyne TWX670 | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalCS-23, Initial Issue (not required by 21A.101, elected to comply by theapplicant)The ce (...) | Project Description: ASR-2009-015-PRD-04-00, Rev. 00Instructions for Installation: ASR-2009-015-IFI-10-00, Rev. 00Instructions for Continued Airworthiness: ASR-2009-015-ICA-11-00, Rev.00Flight Manual Supplement: ASR-2009-015-FMS-05-00, Rev. 00or late (...) | None | 22.04.2010 |
| 10029856 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | | | | 1) M20C, M20E, M20F, M20G, M20J, M20K, M20L, M20M, M20R, M20S | Installation of a Garmin G500 System | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalCS-23, Initial Issue (not required by 21A.101, elected to comply by theapplicant)The ce (...) | Instruction for Installation: ASR-2010-010-IFI-10-00, Rev. 00Instructions for Continued Airworthiness: ASR-2010-010-ICA-11-00, Rev.00Flight Manual Supplement:ASR-2010-010-FMS-05.1-00, Rev. 00ASR-2010-010-FMS-05.2-00, Rev. 00or later revisions of the above listed documents approved by EASA | None | 26.04.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue | |
|--------------|-------------------------------------|--------------|----------------|----------------------|---|--|---|---|--|--|------------|
| 10029947 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | | | | CESSNA 340, 340A | Installation of Thommen AD32 Air Data Display/Altimeter and Avionik Straubing Altitude Preselect System APS4A | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amdt. 1. The certificated noise and/or emissions levels of the original product are (...) | Instructions for Installation: ASR-2009-025-IFI-10-00, Rev. 0. Flight Manual Supplements: ASR-2009-025-FMS-05-00, Rev. 0. Instructions for Continued Airworthiness: ASR-2009-025-ICA-11-00, Rev. 0. Or later revisions of the above listed documents approved by EASA | None | 05.05.2010 | |
| 10030025 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | | | | PA-31-325, PA31-350 PA-31P, PA-31T, PA-31T1, PA-31T2, PA-31T3, PA-31, | Installation of Avdyne TAS 610/620/9900BX Transponder Couplers and Bottom/Top TAS Antennas | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: EASA CS-23, Amendment 1. The requirements for environmental protection and the associated (...) | Instructions for installation: ASR-2010-007-IFI-10-00, Rev. 0. Flight Manual Supplement: ASR-2010-007-FMS-05-00, Rev. 0. Instructions for Continued Airworthiness: ASR-2010-007-ICA-11-00, Rev. 0. Or later revisions of the above listed documents approved by EASA | Restricted to aircraft which have Avdyne EX500/600 MFD, Garmin GDU620PFD, Garmin GNS430(A)W or Garmin GNS530(A)W system installed in order to display the information. | 12.05.2010 | |
| 10030404 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | | | | EG 120 B | Panel Modification including position change of Apollo MX20 (Garmin GXM200), HSI Indicator KI 525 and installation of new Attitude Indicator and Moving Terrain MT-Vision MFD. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 27 Amd. 02. The requirements for environmental protection and the associated certificate (...) | Definition and Installation in accordance with Avionik Straubing GmbH Master Documents List ASR-2010-005-MDL -01-00td16/03/10- Avionik Straubing GmbH Installation Instruction ASR-2010-005-IFI -10-00 Inspection and Maintenance in accordance with: Av (...) | None | 15.06.2010 | |
| 10031180 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | | REV. 1 | | | 1. PA-31, PA-31-325, PA-31-350 2. PA-31P, PA-31T, PA-31T1 2. PA-31T2 | Dual/Single installation of Garmin GNS430(A)W / GNS530(A)W with GPS Antenna Garmin GA35 and Roll Steering Converter DAC GDC31. | CS-23 | Installation must be in accordance with Master Document List Engineering Order No. ASR 2008-163-MDL-01-00 Amend. 00 dated 24. March 2010 or later Approved Revisions. Operation must be in accordance with Flight Manual Supplement No. ASR2008-163-FMS-05.1-0 (...) | none | 13.09.2010 |
| 10032176 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | | REV. 1 | | | 1. PA-32-260, PA-32-300 1. PA-32R-300, PA-32RT-300 1. PA-32R-301T, PA-32-301 1. PA-32RT-300T, PA-32R-301 2. PA-32-301XTC, PA-32-301FT | Dual/Single installation of Garmin GNS430(A)W / GNS530(A)W | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: EASA CS-23, Amdt. 1. The requirements for environmental protection and the associated certificate (...) | Instruction for Installation: ASR-2010-022-IFI-10-00, Rev. 0. Flight Manual Supplement: ASR-2010-022-FMS-05-00, Rev. 0. Instructions for Continued Airworthiness: ASR-2010-022-ICA-11-00, Rev. 0. Or later revisions of the above listed documents approved by EASA | none | 18.10.2010 |
| 10033937 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | | REV. 1 | | | 3101,3102,3103,3107,3108 3109,3110,3112,3116,3201 3202,3206,3207,3212,3216,3217 | Dual/Single installation of Garmin GNS430(A)W / GNS530(A)W and optional installation of Marker Receiver Honeywell (Bendix/King) KMR675 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: EASA CS-23, Amendment 2 (elected to comply by the applicant, not required by 21A.101) The (...) | Instructions for Installation: „ASR-2010-061-IFI-10-00, Rev. 0. Flight Manual Supplement: „ASR-2010-061-FMS-05-1-00, Rev. 0. ASR-2010-061-FMS-05-2-00, Rev. 0. Instructions for Continued Airworthiness: „ASR-2010-061-ICA-11.1-00, Rev. 0. ASR-2010-061-ICA-1 (...) | none | 24.02.2011 |
| 10034601 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | | | | | 210, 210-5, -5A 210A, B, C, D, E, F, G, H, J 210K, L, M, N, R P210N, R T210F, G, H, J, K, L, M, N, R | Dual/Single installation of Garmin GNS430(A)W / GNS530(A)W. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: EASA CS-23, Amdt. 2 (elected to comply by the applicant, not required by 21A.101) The r (...) | Instructions for Installation: ASR-2010-067-IFI-10-00, Rev. 0. Flight Manual Supplement: ASR-2010-067-FMS-05-00, Rev. 0. Instructions for Continued Airworthiness: ASR-2010-067-ICA-11-00, Rev. 0. Or later revisions of the above listed documents approved by EASA | None. | 14.04.2011 |
| 10034991 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | | | | | SEE EASA APPROVED MODEL LIST | Installation of PFD Aspen EFD1000 PRO. | The Certification Basis for the original product remains applicable to this certificate/ approval: CS-23, Amendment 2 (not required by 21A.101) The requirements for environmental protection and the associated certificated noise and/or emissions levels (...) | Flight Manual Supplement: ASR-2011-020-FMS-05-00 Aircraft Model List: ASR-2011-020-AML-24-00 Instructions for Installation: ASR-2011-020-IFI-10-00 Instructions for Continued Airworthiness: ASR-2011-020-ICA-11-00 or later revisions of the above listed documents approved by EASA. | None. | 18.05.2011 |
| 10035162 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | | | | | PA-28-150, -160, -235, -201T PA-28-180, -181, -161, -140 PA-28-236, -151 PA-28R-180, -200, -201, -201T PA-28RT-201, -201T | Installation of dual GPS/NAV/COMM Garmin GNS430/530(A)W. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2 (not required by 21A.101) The requirements for environmental protection (...) | Flight Manual Supplement: ASR-2010-058-FMS-05-00, Amendment 00 Project Description: ASR-2010-058-PRD-04-00, Amendment 00 Instructions for Installation: ASR-2010-058-IFI-10-00, Amendment 00 Instructions for Continued Airworthiness: ASR-2010-058-ICA-11-00 (...) | None. | 31.05.2011 |
| 10035456 | AVIONIK STRAUBING ENTWICKLUNGS GmbH | | | | | F406 | Cockpit Upgrade basing on dual Garmin G600 and GNS430W | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: EASA CS-23, Amdt. 1. The requirements for environmental protection and the associated certificate (...) | Project Description: ASR-2011-033-PRD-04-00, Rev. 0. Instruction for Installation: ASR-2011-033-IFI-10-00, Rev. 0. Instructions for Continued Airworthiness: ASR-2011-033-ICA-11-00<(->->Rev. 0. Flight Manual Supplement ASR-2011-033-FMS-05-00, Rev. 0 (...) | Only applicable for Aircraft S/N 097 and up | 24.06.2011 |
| 10036102 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | REV. 1 | | | DA 40, 40D, 40F | Installation of PFD Aspen EFD1000 PRO. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2. The requirements for environmental protection and the associated certificate (...) | Flight Manual Supplement ASR-2010-069-FMS-05-00, Rev. 0. Or later revisions of the above listed documents approved by EASA. Instructions for Installation: ASR-2010-069-IFI-10-00, Rev. 0. Instructions for Continued Airworthiness: ASR-2010-069-ICA-11-00, Rev. 0. | None. | 17.08.2011 |
| 10036149 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | | SR20, SR22 | Installation of a Primary Function Display Avdyne EXP5000 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2. The requirements for environmental protection and the associated certificate (...) | Flight Manual Supplement: ASR-2011-028-FMS-05-00 or later revisions of the above listed documents approved by EASA. Instructions for Installation: ASR-2011-028-IFI-10-00 Instructions for Continued Airworthiness: ASR-2011-028-ICA-11-00 | The aircraft must have been already equipped with suitable GPS system. | 16.08.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|--------------|----------------|----------------------|--|--|--|---|---|---------------|
| 10036230 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | 501, 551 | Installation of a TAS System L3 Communication SKY899 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2. The requirements for environmental protection and the associated certificate (...) | Flight Manual Supplement: ASR-2011-009-FMS-05-00 Rev. 00 or later revisions of the above listed documents approved by EASA. Instructions for Installation: ASR-2011-009-IFI-10-00 Rev. 00. Instructions for Continued Airworthiness: ASR-2011-009-ICA-11-00 Rev. 00 | The aircraft must have been already equipped with a suitable Multifunction Display. | 23.08.2011 |
| 10036231 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | 208, 208B | Installation of a Satellite Antenna Chelton HGA-7000 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2. The requirements for environmental protection and the associated certificate (...) | Flight Manual Supplement: ASR-2011-012-FMS-05-00 Rev. 00 or later revisions of the above listed documents approved by EASA. Instructions for Installation: ASR-2011-012-IFI-10-00 Rev. 00. Instructions for Continued Airworthiness: ASR-2011-012-ICA-11-00 Rev. 00 | none | 23.08.2011 |
| 10036587 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | REV. 1 | | 1. PA-31-325, PA-31-350, 1. PA-31, PA-31-300, 2. PA-31P, PA-31P-350, 2. PA-31T, PA-31T1, 2. PA-31T2, PA-31T3 | Installation of one or more Avionik Straubing Heater Control Unit(s) HCU200. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2. The requirements for environmental protection and the associated certificate (...) | Flight Manual Supplement ASR-2010-027-FMS-05-00, Rev. 00. Instructions for Installation ASR-2010-027-IFI-10-00, Rev. 00. Instructions for Continued Airworthiness ASR-2010-027-ICA-11-00 (-, -) Rev. 00 or later revisions of the above listed documents approved by EASA. | None. | 04.10.2011 |
| 10036677 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | 60, A60, B60 | Installation of a Traffic Advisory System Avidyne TAS 605/615/620 (-, -) Ryan 9900BX | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2. The requirements for environmental protection and the associated certificate (...) | Flight Manual Supplement ASR-2011-016-FMS-05-00, Rev. 00. Instructions for Installation ASR-2011-016-IFI-10-00, Rev. 00. Instructions for Continued Airworthiness ASR-2011-016-ICA-11-00, Rev. 00 or later revisions of the above listed documents approved by EASA. | None | 28.09.2011 |
| 10036784 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | BEECH 60, A60, AND B60 | Installation of Garmin G600 System with Synthetic Vision Technology (SVT). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Avionik Straubing Engineering Order ref ASR-2011-015, Amendment 00 dated 4/4/11. Operation in accordance with Avionik Straubing Flight Manual Supplement ref ASR-2010-015-FMS-05-00, Amendment 00, dated 4/4/11. Instructions (...) | None. | 06.10.2011 |
| 10036858 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | 60, A60, B60 | Installation of a Satellite Data Link System Garmin GSR56. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2. The requirements for environmental protection and the associated certificate (...) | Flight Manual Supplement: ASR-2011-051-FMS-05-00, Rev. 00. Instructions for Installation: ASR-2011-051-IFI-10-00, Rev. 00. Instructions for Continued Airworthiness: ASR-2011-051-ICA-11-00, Rev. 00 or later revisions of the above listed documents approved by EASA. | None. | 12.10.2011 |
| 10037583 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | 1. PA-31-350 1. PA-31, PA-31-300, PA-31-325 2. PA-31P, PA-31P-350 2. PA-31T, PA-31T1, PA-31T2 2. PA-31T3 | Installation of Garmin G600 System with Synthetic Vision (SVT) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2. The requirements for environmental protection and the associated certificate (...) | Flight Manual Supplement G600: ASR-2011-033-FMS-05-1-00, Rev. 00. Flight Manual Supplement EPU6AS: ASR-2011-033-FMS-05-2-00, Rev. 00. Instruction for Installation Garmin G600: ASR-2011-033-IFI-10-1-00, Rev. 00. Instruction for Installation Garmin EPU6A (...) | None | 08.12.2011 |
| 10037817 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | 1. M20, M20A, M20B, M20C, M20D 1. M20E, M20F, M20G, M20J 1. M20K, M20L, M20S 2. M20M, M20R | Dual/Single Installation of Garmin GPS/COMM/NAV GNS430(A)/W/GNS530(A)/W | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2 (not required by 21A.101). The requirements for environmental protection (...) | Flight Manual Supplement ASR-2011-054-FMS-05-00 Rev. 00 or later revisions of the above listed documents approved by EASA. Instructions for Installation ASR-2011-054-IFI-10-00 Rev. 0. Instructions for Continued Airworthiness ASR-2011-054-ICA-11-00 Rev. 0 | None | 22.12.2011 |
| 10038943 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | SR20 SR22 | Installation of PFD Aspen EFD1000 PRO. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2 (not required by 21A.101). The requirements for environmental protection (...) | Flight Manual Supplement ASR-2011-060-FMS-05-00 Rev. 00 or later revisions of the above listed documents approved by EASA. Instructions for Installation ASR-2011-060-IFI-10-00 Rev. 00. Instructions for Continued Airworthiness ASR-2011-060-ICA-11-00 Rev. 00 | None. | 23.03.2012 |
| 10040468 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | PA-42, PA-42-720, PA-42-1000 | Installation of Garmin G600 System with Synthetic Vision (SVT) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2 (not required by 21A.101). The requirements for environmental protection (...) | Flight Manual Supplement G600: ASR-2012-016-FMS-05-00, Rev. 00 or later revisions of the above listed documents approved by EASA. Instructions for Installation: ASR-2012-016-IFI-10-00, Rev. 00. Instructions for Continued Airworthiness: ASR-2012-016-ICA-11-00, Rev. 00 | none | 05.07.2012 |
| 10041049 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | DA40, DA40D, DA40F | Dual/Single Installation of Garmin GPS/COMM/NAV GNS430W / GNS530W. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2. The requirements for environmental protection and the associated certificate (...) | Flight Manual Supplement ASR-2012-052-FMS-05-00, Rev. 00 or later revisions of the above listed documents approved by EASA. Instructions for Installation ASR-2012-052-IFI-10-00, Rev. 00. Instructions for Continued Airworthiness ASR-2012-052-ICA-11-00, Rev. 00. | None. | 17.08.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|----------------|----------------|----------------------|--|--|--|---|---|---------------|
| 10041441 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | REV. 1 | | 650 | Dual Installation of Flight management System Universal UNS-1Lw | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Avionik Straubing Master Document List: ref ASR-2012-028-MDL-01-01, rev01, date 20 September 2012/Avionik Straubing AFM Supplement: ref ASR-2012-028-FMS-05-01, rev 01, date 20 September 2012/Avionik Straubing Instructions for Continued Airworthiness, r (...) | none | 20.09.2012 |
| 10042720 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | SC7 SERIES 3 | Installation of PFD ASPEN EFD1000 C3 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 3 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement PFD ASR-2012-004-FMS-05-1-00, Rev. 00- Flight Manual Supplement MFD ASR-2012-004-FMS-05-2-00, Rev. 00 or later revisions of the above listed documents approved by EASA - Instructions for Installation ASR-2012-004-IFI-10-00, (...) | None. | 14.12.2012 |
| 10042753 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | 525, 525A | Single Installation of GPS/COMM/NAV System Garmin GTN725/750. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2 (not required by 21A.101) The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement ASR-2012-053-FMS-05-00 Rev. 00 or later revisions of the above listed documents approved by EASA. Instructions for Installation ASR-2012-053-IFI-10-00 Rev. 00 Instructions for Continued Airworthiness ASR-2012-053-ICA-11-00 Rev. 00 | None. | 17.12.2012 |
| 10042880 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | B200, B200C, B200GT, B200CGT | Installation of Flight Management System Universal UNS-1Lw | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2 (not required by 21A.201) The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement ASR-2012-057-FMS-05-00 Rev. 00 or later revisions of the above listed documents approved by EASA Instructions for Installation ASR-2012-057-IFI-10-00 Rev. 00 Instructions for Continued Airworthiness ASR-2012-057-ICA-11-00 Rev. 00 | None | 21.12.2012 |
| 10043138 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | EASA.A.S.03733 | | | PA34-200, PA34-200T AND PA34-200T | Dual installation of Garmin GNS430W/ GNS430W/ GNS 430/ GNS530W/GNS530AW/ GNS530. | | Installation must be in accordance with Master Document List Engineering Order No. ASR2008-125-MDL-01-00 Amend. 00 dated 3-- Jan. 2009 or later Approved Revisions. Operation must be in accordance with Flight Manual Supplement No. ASR2008-125-FMS-05-1-00 (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The (...) | 02.06.2009 |
| 10043270 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | MBB-BK 117 A-1, A-3, A-4 MBB-BK 117 B-1, B-2, C-1, C-2 | Installation of Cockpit Voice Recorder L-3 FA2100CVR as defined in ASR-2012-093-PRD-04-00. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ASR-2012-093-TCP-06-00 dated 09/11/2012/ASR-2012-093-PRD-04-00 dated 14/11/2012/ASR-2012-093-COR-07-00 dated 10/01/2013/ASR-2012-093-MDL-01-00 dated 10/01/2013 | None | 17.01.2013 |
| 10044173 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | 401/401A/401B/402/402A/402B/402C/411/411A/414/414A/421/421A/421B/421C/425 | Installation of Garmin GTS820 with GSA65 and Dual Garmin GA58, Garmin GDL89, DAGDC31, Garmin GA36, Garmin GA37 and Comant CI 490-1 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23 Amendment 3 for following paragraphs: 23.301, 303, 305, 307, 365, 561, 571, 575, 603, 605, 607, 609, 611, 613, 771, 773, 843, 863, 867, 1182, 1301, 1309, 1321, 1322, 1 | Flight Manual Supplement Garmin GTS820 ASR-2013-015-FMS-05-1-00, Rev. 0-Flight Manual Supplement GDC31 ASR-2013-015-FMS-05-2-00, Rev. 0 or later revisions of the above listed documents approved by EASA - Instructions for Installation ASR-2013-015-IFI-10-00, (...) | None | 22.03.2013 |
| 10045024 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | REV. 1 | | AA-5, AA-5A, AA-5B, AG-5B | Installation of Garmin G500 System | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, Amendment 3 (not required by 21A.101) 23.303, 23.305, 23.561, 23.575, 603, 605, 607, 609, 611, 613, 771, 773, 777, 863, 867, 95, 4, 1 (...) | Flight Manual Supplement G500 ASR-2013-004-FMS-05-1-00 Rev. 00- Flight Manual Supplement GDL88 ASR-2013-004-FMS-05-2-00 Rev. 00 or later revisions of the above listed documents approved by EASA - Instructions for Installation ASR-2013-004-IFI-10-00, (...) | A suitable Garmin GPS System must have been already installed. This STC includes the installation of Garmin's Synthetic Vision (SVT) Option and the optional installation of a Data Link Sensor System Garmin GDL88 and/or an Avionik Straubing Altitude Presse (...) | 31.05.2013 |
| 10045230 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | 525, 525A, 525B | Installation of Dual Garmin GTN650/GTN 750 Systems | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, Amendment 3 23.301, 23.303, 23.305, 23.307, 23.309, 23.561, 23.571, 23.575, 23.603, 23.605, 23.607, 23.609, 23.611, (...) | Flight Manual Supplement ASR-2013-023-FMS-05-01, Amdt. 01 or later revisions of the above listed documents approved by EASA. Instructions for Installation ASR-2013-023-IFI-10-01, Amdt. 01, or later approved revisions Instructions for Continued Airworthiness (...) | None | 10.06.2013 |
| 10045310 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | 1. SA226-AT, -T, -T(B) 1. SA227-AT, -TT 1. SA226-AT, -T 2. SA226-TC 2. SA227-AC, -BC, -PC 3. SA227-CC, -DC, | Single Installation GPS/COMM/NAV System Garmin GTN625/635/650/725/750 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, Amendment 3 (not required by 21A.101) 23.303, 23.305, 23.561, 23.575, 23.603, 23.605, 23.607, 23.609, 23.611, 23.613, (...) | Flight Manual Supplement ASR-2013-049-FMS-05-00, Rev. 00 or later revisions of the above listed documents approved by EASA Instructions for Installation ASR-2013-049-IFI-10-00, Rev. 00 or later approved revisions Instructions for Continued Airworthiness (...) | The aircraft must have been already equipped with a suitable GPS Antenna and a suitable E-HSI. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 14.06.2013 |
| 10045451 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | PC-12, PC-12/45, PC-12/47, PC-12/47E. | Installation of L-3 Communications WX-500 Stormscope with Antenna NY-163 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-23, Amendment 3, 23.301, 303, 305, 307, 365, 561, 571, 575, 603, 605, 6 (...) | Flight Manual Supplement ASR-2013-001-FMS-05-00, Amdt. 00 or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 26.06.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|----------------|----------------|----------------------|------------------------------|--|--|--|--|---------------|
| 10045531 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | 501, 551 | Installation of TAS System Garmin GTS820 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, Amendment 3301, 303, 305, 307, 365, 561, 571, 575, 603, 605, 607, 609, 611, 613, 771, 773, 843, 863, 867, 1182, 130 (...) | Flight Manual ASR-2013-019-FMS-05-00, Rev. 00 or later revisions of the above listed documents approved by EASA. Instructions for Installation ASR-2013-019-IFI-10-00, Rev. 00, or later approved revision instructions for Continued Airworthiness ASR-2013-0 (...) | The aircraft must have been already equipped with a suitable display. | 03.07.2013 |
| 10045666 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | | | B200, B200C, B200GT, B200CGT | Installation of Universal Avionics UNS-1E NCU, Config Module and SSDTU. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Avionics Straubing modification ref ASR-2013-051 Am 00 dated 15/05/2013; Operate in accordance with Avionics Straubing Flight Manual Supplement ref ASR-2013-051-FMS-05-00 dated 15/05/2013; Maintain in accordance with Avion (...) | None | 10.07.2013 |
| 10045765 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | REV. 1 | | 501 AND 551 | Single/Dual Installation of GPS/COMM/NAV System Garmin GTN650/750 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-23, Amendment 3 (not required by 21.A.101) 301, 303, 305, 307, 365, 56 (...) | Flight Manual Supplement ASR-2013-018-FMS-05-00, Rev. 00 or later revisions of the above listed documents approved by EASA. Instructions for Installation ASR-2013-018-IFI-10-00, Rev. 00 or later approved revision instructions for Continued Airworthiness (...) | The aircraft must have been already equipped with (a) suitable E-HSI(s). Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or re (...) | 22.07.2013 |
| 10045766 | AVIONIK STRAUBING ENTWICKLUNGS GMBH | | REV. 1 | | 501 AND 551 | Installation of dual E-HSI Sandel SN4500 and single E-ADI Sandel SN4550 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-23, Amendment 3 (not required by 21.A.101) 307, 561, 575, 603, 605, 60 (...) | Flight Manual Supplement: ASR-2013-021-FMS-05-1-00, Rev. 00- ASR-2013-021-FMS-05-2-00, Rev. 00 or later revisions of the above listed documents approved by EASA. Instructions for Installation ASR-2013-021-IFI-10-00, Rev. 00 or later approved revision (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 22.07.2013 |
| 10014810 | AVIONITEC AG | EASA.A.S.03015 | | | CESSNA 501 | Replacement of XPDR to GARMIN GTX-330D Diversity Mode S and change of GPS interface | | Master Document List 063269-017, latest approved Revision Avionitec AFM-S Doc. 063269-012 from 14. November 2007, or later EASA approved Revision Avionitec AFM-S Doc. 063269-011 from 7. November 2007, or later EASA approved Revision Avionitec Instruction (...) | | 29.01.2008 |
| 10014846 | AVIONITEC AG | EASA.A.S.03067 | | | PIPER PA-31T1 | Replacing of existing avionics by newer generation of avionics in accordance with Unit change overview Nr. 063247-050. | | Installation must be in accordance with Avionitec Master Document List No. 63247-017 Rev. 1 dated 19 December 2007 or later approved revisions. Operation must be in accordance with Avionitec Airplane Flight Manual Supplement No. 063247-011 and 063247- (...) | Limited to S/N 31T-7904014 | 17.01.2008 |
| 10014858 | AVIONITEC AG | EASA.A.S.03082 | | | PIPER PA-60-601P | Avionics retrofit: Out. Radar system Bendix RDR-150; VHF-COM King KY-196; VHF-NAV King KN-53, Indicators KI-252A, KI-204, and KI-226; ADF system King KR-87 incl. Antenna KA-44B, obsolete ADF antenna Collins ANT-650A and adapter ADA-650; Moving map Even (...) | | AFM supplement No. 073301-011 Garmin GNS-530AFM supplement No. 073301-012 Bendix/King KMH-880AFM supplement No. 073301-013 Sandel SN-3500 | Reg.: HB-LIN S/N: 571-7963248/ | 08.10.2007 |
| 10014864 | AVIONITEC AG | EASA.A.S.03088 | | | MOONEY M20J | Installation of replaced Avionics according CP No. 073372 | | According to Avionitec Master Document List Doc. No. 073372-017 Rev. 1, or later Revisions To be operated in accordance with approved Avionitec AFM-S Document No. 073372-011, Original issue from 23.05.2008, or later approved Revisions To be maintained (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 21.11.2008 |
| 10014916 | AVIONITEC AG | EASA.A.S.03162 | | | PIPER PA-28R-201T | Replacing of existing avionics by newer generation of avionics in accordance with Unit change overview Nr. 073528-050, Rev 0, dated 11.06.07. | | Installation must be in accordance with Avionitec Master Document List No. 073528-017 Rev. 0 dated 20. Dec. 2007 or later Approved Revisions. Operation must be in accordance with Avionitec Airplane Flight Manual Supplement No. 073528-011 dated 06. Jul (...) | Limited to S/N T21059922. | 08.05.2008 |
| 10014986 | AVIONITEC AG | EASA.A.S.03278 | REV. 1 | | | Avionics Retrofit: Out. King Mode A/C Transponder KI-76A, King Audio Panel KMA-24, Autopilot Badin Crozet In: Garmin Mode S Transponder GTX-330, Garmin Audio Panel GMA-340, Autopilot S-Tec System 55X. | | Master Document List No. 042220-MDLAFM Supplement No. 042220-AFMS. | Reg.: HB-DCE, S/N 420. | 06.09.2007 |
| 10015064 | AVIONITEC AG | EASA.A.S.03390 | | | CESSNA T210L | Replacing of existing avionics by newer generation of avionics in accordance with Unit change overview Nr. 073775-050, Rev 0, dated 05.12.07. | | Installation must be in accordance with Avionitec Master Document List No. 073775-017 Rev. 1 dated 14. Feb. 2008 or later Approved Revisions. Operation must be in accordance with Avionitec Airplane Flight Manual Supplement No. 073775-011 dated 08. Jan (...) | Limited to S/N T21059922. | 27.02.2008 |
| 10015104 | AVIONITEC AG | EASA.A.S.03447 | | | BEECH F90 | Avionics replacing: Garmin GTX330D XPDR and Weather Radar System RDR 2000. | | Installation must be in accordance with Avionitec Master Document List No. 073839-017 Rev. 0 dated 19. Mar. 2008 or later Approved Revisions. Operation must be in accordance with Avionitec Airplane Flight Manual Supplement No. 073839-011 dated 17. Feb (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 23.09.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------|----------------|----------------|----------------------|--------------------------------|---|--|--|---|---------------|
| 10015163 | AVIONITEC AG | EASA.A.S.03534 | | | PIPER PA-46-350P | Avionics upgrade to GNS 430, Removed Equipment:- KX-165 NAV/COM 2- KI-206 NAV 2 Indicator/Installed Equipment- GNS-430 NAV/COM/GPS 2- GI-106A NAV/GPS Indicator 2- GA-56 GPS 2 Antenna | | - Master Document List, Doc. No. 073723-017, Rev. 1- AFMS, Doc. No. 073723-011, Rev. 1 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.Reg. - HB-PPH, S/N: 4636045 | 30.06.2008 |
| 10015191 | AVIONITEC AG | EASA.A.S.03575 | | | COMMANDER AIRCRAFT RC114 | Avionics Retrofit/Removed Equipment:- KMA 20 Audio Panel- Isocom Intercom- Dual KX 155 COM/NAV- KI 203 NAV 2 Indicator-GPS 1, Garmin GPS 100 incl. Antenna- OEM vacuum Directional Gyro- KT 76A Transponder- Narco ELT 10New installed Equipment:- GMA 3 (...) | | - Master Document List, Doc. No. PE8100-017, Rev. 1 or latest approved Revision- AFMS Dual GNS 430, Doc. No. PE8100-011, Original Issue or latest approved Revision- AFMS SN3500 HIS, Doc. No. PE8100-012, Original Issue or latest approved Revision- AF (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 23.12.2008 |
| 10015278 | AVIONITEC AG | EASA.A.S.03794 | | | PIPER PA-46-350P (MALIBU MIRAG | Installation Weather avoidance systems and TAS and ME406 ELT by installation of:- Avidyne MFD EX 500- Avidyne Traffic Advisory system TAS 610- Avidyne Tactical Weather detection System TWX 670- Avidyne Iridium Data Link Transceiver MLX 770- Artex ME406 ELT | | Master Document List: P181151-070; Instruction for continued airworthiness: P181151-015; Declaration of Compliance: P181151-060; AFMS EX500: P181151-014; AFMS MLX770: P181151-012; AFMS TAS 610: P181151-011; AFMS TWX670: P181151-013 | Reg.: HB-POY, SN: 4622090 | 10.03.2009 |
| 10033155 | AVIONITEC AG | | | | T303 | Avionics Retrofit/Removed:-Garmin GNS 430 NAV/COM/GPS-King KX 165 NAV/COM with King KI 206 Indicator-ARC RTA 476A DME- ARC R-446A ADF with ARC IN 346A Indicator/Installed:-Garmin GNS530W NAV/COM/GPS (upgrade from a Garmin GNS 430)-Garmin GNS430W NAV/COM/GPS with Garmin GI 106A Ind.-King KNG2A DME | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List: Doc. No. P191405-017, Rev. 0-Instructions for Continued Airworthiness: Doc. No. P191405-015, Rev.0- Airplane Flight Manual Supplement: Doc. No. P191405-011, Rev. 1or later revisions of the above listed documents approved by EASA | Limited to: HB-LUV, S/N T30300058 | 21.12.2010 |
| 10034285 | AVIONITEC AG | | REV. 1 | | PA-46-310P | Avionics Retrofit/Removed Equipment:-Dual KX 165 NAV/COM System-,WX 1000 Stormscope System-,Argus 5000 Moving Map-,KMA 24 Audio Panel-,PM 2000 IntercomNew installed Equipment:-,GNS 530W NAV/COM/GPS System 1-,GNS 430W NAV/COM/GPS System 2-, (...) | FAR 23, Amdt: 25 (tick one of the boxes below if applicable)The Certification Basis for the original product remains applicable to this certificate/ approval.Derogation to latest amendment (Part 21A.101): Do not materially contribute to safety. | - Master Document List, Doc. No. P101470-017, Rev. 0, 17.02.2011- Declaration of Compliance, Doc. No. P101470-60, Rev. 0, 17.02.2011- ICA, Doc. No. P101470-015, Rev. 0, 17.02.2011- AFMS Dual GNS 530W/430W, Doc. No. P101470-011, Rev. 0, 17.02.2011- A (...) | S/N: HB-PJW | 25.03.2011 |
| 10034590 | AVIONITEC AG | | | | M20K | Avionics Retrofit/Removed: -, King KX 165 NAV/COM with King KI 206 Indicator- King KR 87 ADF with King KI 229 Indicator- Garmin GNS530 NAV/COM/GPS/Installed: -, Garmin 530W NAV/COM/GPS- Garmin 430W NAV/COM/GPS with GI 106A Indicator/Vertically guided (...) | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List: No. P191407-017, date 11.04.2011Airplane Flight Manual Supplement: No P191407-11, date 11.02.2010Instruction for continued airworthiness: No. P191407-015, date11.02.2010or later revisions of the above listed documents approved by EASA. | S/N: 25-1170 HB-DHG | 13.04.2011 |
| 10037582 | AVIONITEC Ltd. | | | | PA-28-236 | IFR Avionics upgrade/Removed:- King KY 157A COM- King KMA 20Audio Panel- King KI 53 NAV Receiver- King KI 203 NAV Indicator- David Clark Isocom Intercom/Installed:- Garmin GNS530W GPS/COM/NAV with GA35 antenna- Garmin GMA 347 Audio Panel- King KR 87 A (...) | FAR 23The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List: Doc. No. P101488-017, dated 26.08.2011Instructions for continued airworthiness: P101488-015, dated 26.08.2011Airplane Flight Manual Supplement: P101488-011, dated 26.08.2011 | S/N: 28-8011011 HB-PEN | 08.12.2011 |
| 10038996 | AVIONITEC Ltd. | | REV. 1 | | BEECH 35-C33A | Upgrade Garmin GNS430, repl. NAV/COM/Removed:-Garmin GNS430 System-King KX 155 NAV/COM system-King KR 85 ADF System/Installed:-Dual Garmin GNS430W System-King KR 87 ADF System | FAR 23, Amendment 7The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original prod (...) | -Master Document List: Doc. No. 1098-017, Rev. 0-Airplane Flight Manual Supplement: Doc. No. 1098-011, Rev. 1-Instructions for Continued Airworthiness: Doc. No. 1098-015, Iss.24.11.2011or later revisions of the above listed documents approved by EASA | S/N CE-39, F-BNOI | 05.04.2012 |
| 10043286 | AVIONITEC Ltd. | | | | 335 | Compliance showing to AMC20-28 for certification of existing equipment for RNAV (GNSS) Approaches down to LPV Minima.Compliance showing to AMC20-27, resp. CRI F-10 (Airworthiness Criteria) for RNAV Equipment supporting APV BARO VNAV approaches to LNAV/ (...) | FAR 23<->-> Amdt: 23The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List: Doc. No. 2307-017, dated 14.12.2012Instructions for continued airworthiness: Doc. No. 1193-015, dated08.12.2011Airplane Flight Manual Supplement for dual GNS430W Installation: Doc.No. 1193-011, dated 14.12.2012 | S/N: HB-LHA | 17.01.2013 |
| 10044281 | AVIONITEC Ltd. | | | | P210N | Removed:- Sigtronics SPA400 Intercom-King KMA 24 Audio panel- King KX 165 NAV/COM- King KLN90B GPS/Installed:-Garmin GMA 350 Intercom/Audio panel-Garmin GNS 430W NAV/COM/GPS- L-3 WX-500 StormscopeCompliance showing to AMC20-28 that the installed (...) | FAR 23<->-> Amdt: 16The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List: Doc. No. 2259-017, dated 20.02.13Instructions for continued airworthiness: Doc. No. 2259-015, dated30.11.12Airplane Flight Manual Supplement: Doc. No. 2259-011, dated 20.02.2013or later revisions of the above listed documents approved by EASA | S/N: P210-0134<->-> HB-TCSThe approval holder shall fulfill the obligations of Part 21 §21.A.118Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed m (...) | 02.04.2013 |
| 10045732 | AVIONITEC Ltd. | | | | PA-34-200 | Dual GNS upgrade to WAAS Units incl. LPV Certification/Compliance showing to AMC20-28 that the installed equipment is suitable for RNAV (GNSS) Approaches down to LPV Minima.Compliance showing to AMC20-27, resp. CRI F-10 (Airworthiness Criteria) for RNAV (...) | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List: 1418-17, dated 28.05.2013Instructions for continued airworthiness: 1418-015, dated 15.04.2013Airplane Flight Manual Supplement: 1418-011, dated 23.02.2013or later revisions of the above listed documents approved by EASA. | S/N: 34-7350164 HB-LQJThe approval holder shall fulfill the obligations of Part 21 §21.A.118.Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed m (...) | 15.07.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|-------------------|----------------|----------------------|---|---|--|---|--|---------------|
| 10037702 | AZRA LLC | EASA.IM.A.S.0116 | REV. 2 | | 737-200, 737-200C, 737-300 737-400, 737-500 | Modification of Wing Inboard & Outboard Aft Trailing Edge Flaps. Revision 2 of the STC with reference EASA.IM.A.S.0116 has been issued as a result of its transfer from AeroTech Services Inc (ATS) to AZRA LLC DBA AeroTech Services. | The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval. | - AeroTech Services Inc, Master Drawing List No 52-0001 Revision C dated May 20 2003 (for Model 737-200, -200C and 300 Series Airplanes) - AeroTech Services, Inc Master Drawing List No 52-0001, Revision E, dated January 6 (for Model 737-400 and -500 Ser (...)) | Scheduled inspections and repairs are required in accordance with the following documents: 1. Chapter 05 of the "737-200, -200C Aircraft Maintenance Manual Supplement" Report No. ATSS2-0008, Issue 4, dated 6/30/05, for 737-200 and -200C, with this STC modification, (and Issue 4 of ATSS2-0008 superseded) | 22.12.2011 |
| 10014133 | AZUR AERO ASSISTANCE | EASA.A.S.01244 | | | CESSNA 500, CITATION | Installation of Radar RDR 2000 Bendix/King with radar screen KMD 540/850 Bendix/King in accordance with MDL AAA 08/2004 ed 1 Rev 1 dated 10/10/05 | | AFM Supplemental AAA 08/2004 Ed 1 dated 20/10/2004 | This STC is limited to MSN 308. | 10.11.2005 |
| 10014134 | AZUR AERO ASSISTANCE | EASA.A.S.01245 | | | CESSNA 500, CITATION | Installation of Mode S Transponder GTX 330 Garmin in accordance with MDL AAA 07/2004 Ed 1 Rev. 1 dated 10/10/2005 | | AFM Supplement AAA 07/2004 Ed 1 dated 19/10/2004 | This STC is limited to MSN 308. | 10.11.2005 |
| 10014135 | AZUR AERO ASSISTANCE | EASA.A.S.01246 | | | CESSNA 500, CITATION | Installation of GPS GNS 430 GARMIN in accordance with MDL AAA 06/2004 Ed 1 Rev 1 dated 10/10/2005 | | AFM Supplement AAA 06/2004 Ed 1 dated 19/10/2004 | This STC is limited to MSN 308. | 10.11.2005 |
| 10014136 | AZUR AERO ASSISTANCE | EASA.A.S.01247 | | | CESSNA 500, CITATION | Installation of GPS GNS 530 BRNAV GARMIN in accordance with MDL AAA 05/2004 Ed 1 Rev 1 dated 10/10/2005 | | AFM Supplement AAA 05/2004 Ed 1 dated 19/10/2004 | This STC is limited to MSN 308 | 10.11.2005 |
| 10016328 | B&B PARTNERS | EASA.IM.A.S.02637 | | FAA STC ST03483AT | | Installation of a portable beverage dispensing system, P/N B43-FA 001 | | - FAA STC ST03483AT dated 24 September 2007 - B&b Partners Assembly Drawing B43-FA, dated 21 December 2006 - Master Document List No. 6009, dated 14 August 2007 - Instructions for Continued Airworthiness for P/N B43-FA-001, Rev. A dated 9 Au (...) | 1. This STC is applicable to the aircraft types as listed in the Approved Model List for FAA STC ST03483AT2. This STC is to be installed in accordance with the Galley Interface Criteria defined in Drawing No. 1178.3. The limitations as defined in FAA STC ST03483AT are applicable | 06.02.2008 |
| 10016351 | B&C SPECIALITY PRODUCTS | EASA.IM.A.S.02671 | | FAA STC SA00846W1 | CESSNA 210L CESSNA 210M CESSNA 210N CESSNA T210L CESSNA T210M CESSNA T210N | Addition of Vacuum Pump drive alternator and regulator with associated controls. | | | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 28.11.2007 |
| 10033488 | B/E AEROSPACE, INC. | | | FAA STC ST14379LA-T | 777-300, 777-300ER | Installation of B/E Aerospace Pantry Air Chiller P/N 470-1 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: Special Condition CRI H-01 EWIS The certified noise and/ or emissions levels of the or (...) | Master drawing List ECS470-42 revision A dated June 3, 2010 Instruction for Continued Airworthiness ECS470-43 revision C dated July 9, 2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | Not Applicable | 25.01.2011 |
| 10036542 | B/E AEROSPACE, Inc. | | | FAA STC ST02876CH | B777-200, B777-200LR B777-300, B777-300ER B777F | Installation of B/E Aerospace, Inc., Vacuum Toilet Assembly Retrofit Kit | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-1 EWIS ICA | Installation Instructions AMM 38-30-03 Rev. 3 dtd. 10 May 2011 Instructions for Continued Airworthiness 38-30-04 Rev. 3 dtd. 2 June 2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 19.09.2011 |
| 10036558 | B/E AEROSPACE, Inc. | | | FAA STC ST02876CH | B737-600/700/800/900 | Installation of Vacuum Toilet Assembly Retrofit Kit P/N 4100390-003. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-1 EWIS ICA The certified noise and/ or emissions levels of the original product (...) | Component Certification Report Vacuum Toilet Assembly Retrofit Kit CCR4100390-003 rev F released on 15-09-2011; Master Data List MDL 4100390-001 rev A- Master Data List MDL 4100390-003 rev B or later revisions of the above listed documents approved (...) | None. | 19.09.2011 |
| 10036659 | B/E AEROSPACE, Inc. | | | FAA STC ST02876CH | B747-400 B747-400F | Installation of Vacuum Toilet Assembly Retrofit Kit P/N 4100390-003 Validation FAA STC ST02876CH, .. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS The requirement (...) | AMM Supplement 38-30-06-02 Instructions for Continued Airworthiness (ICA) and EWIS ICA source document, dated April 15, 2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None. | 27.09.2011 |
| 10040968 | B/E AEROSPACE, Inc. | | | FAA STC ST02876CH | B767-200/-300/-300F/-400ER | Installation of BE Aerospace Inc. Vacuum Toilet Assembly Retrofit Kit | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this approval. | BE Aerospace Installation Instructions, AMM 38-30-07, Revision 01, 11/08/2011. BE Aerospace ICA document, AMM 38-30-08, Revision 01, 11/04/2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 09.08.2012 |
| 10016917 | BAE SYSTEMS AEROSPACE & DEFENSE | EASA.IM.R.S.01302 | REV. 1 | FAA STC SR01906LA | BK117 C-2 | Installation of the Folding Utility Seat in Multiple-Place Seat Arrangements. The Utility Seating System, as installed in the MBB BK117 C-2 helicopter, multiple-place utility seating system consisting of energy-absorbing Utility Installation of the Fold (...) | The Certification Basis for the original product remains applicable to this certificate/ approval | RFMS Document No. TD-05174 Rev. F, FAA approved 01.10.2009 - Master Drawing List (MDL) TD-05175 Rev. R, FAA approved 31.08.2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 14.12.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|----------------------|--|---|---------------------|---|---|---------------|
| 10015325 | BALLISTIC RECOVERY SYSTEMS INC | EASA.IM.A.S.00280 | | FAA STC SA01999CH | 182G, 182H, 182J, 182K, 182L 182M, 182N, 182P, 182Q, 182R, 182S, 182T, T182, T182T | BRS-182 Parachute Recovery System FAA STC SA01999CH. In accordance with Ballistics Recovery Systems Inc Drawing No 9005 Revision A dated June 11 2004 or later Approved Revision.Flight Manual Supplement Ballistic Recovery Systems AFM Supplement D (...) | | | 1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated.2. If transfer of this Certificate is requested under Part 21A.116, the (...) | 15.10.2004 |
| 10015952 | BARBARA AND ROBERT WILLIAMS | EASA.IM.A.S.02094 | | SA572CE | CESSNA 150 CESSNA 150A CESSNA 150B CESSNA 150C CESSNA 150D CESSNA 150E CESSNA 150F CESSNA 150G CESSNA 150H CESSNA 150J CESSNA 150K CESSNA 150L CESSNA 150M CESSNA A150K CESSNA A150L CESSNA A150L | Replacement of Continental O-200A with Lycoming 150 or 160hp Model O-320 engines | | Installation, operation and Maintenance in accordance with the documents and data contained in FAA STC SA572CE at latest approved revision | As defined in FAA STC SA572CE Modified engine mounting structure and relocated battery installation must be inspected every 500 flight hours or annually, whichever occurs earlier for any cracking or corrosion and if found rectified with an EASA approved repair scheme | 20.09.2006 |
| 10016380 | BASIC AIRCRAFT PRODUCTS, INC. | EASA.IM.A.S.02704 | | FAA STC ST00702AT | CESSNA 210 | Installation of Basic Type mBAE-14/28 Model 2025 Deployable Air Driven Alternator System | | Basic Aircraft Products, inc. Installation Kit No. 45-2025-OPL, Rev. A, dated 26-09-1994 and Installation Manual No. 80-2025-101, dated 24-06-1994 or later approved revisions. | This approval should not be extended to other aircraft of this model on which other previously approved modifications will produce adverse effect upon the airworthiness of that aircraft. EASA approved Supplemental Airplane Flight Manual or EASA appro (...) | 18.01.2008 |
| 10016497 | BASIC AIRCRAFT PRODUCTS, INC. | EASA.IM.A.S.02859 | | FAA STC SA02747AT | CESSNA 182 | Stand-by Alternator # Installation of Basic Type BAE-14/28 Deployable Air Driven Alternator System. | | The installation will be done in accordance with Basic Aircraft Products document #INSTALLATION MANUAL NO. 80-2625-101A#, issued July 5, 2003 or later FAA approved revision.The applicable AFMS as per Basic Aircraft Products document BAE-14/28 Air Dr (...) | As per FAA STC SA02747AT limitations. | 29.08.2008 |
| 10043069 | BAUMANN FLOATS LLC | EASA.IM.A.S.03190 | | FAA STC SA01491CH | M-5-235C,M-6-235,M-7-235 M-7-235B,MX-7-235,M1-7-235 | Installation of BF2750A amphibious float installation | | According to Baumann Master Document / Drawing List Doc. No. BF2750A-02Rev. F, or later FAA approved Revisions To be operated in accordance with Baumann AFM-S Documents No. BF2750A,rev. E, or later FAA approved Revisions To be maintained i.a.w. JCA Do (...) | none | 27.05.2009 |
| 10014838 | BAYO S.A.S. | EASA.A.S.03055 | | | MOONEY M20K | Modification of radio electric installation for B-RNAV capacity with AP/FD coupling. | | Modification file No. STC 18/2007 Flight Manual Supplement No. STC 18/2007 dated 6 March 2007. | Modification limited to serial number 250508AFMS must be onboard for any flight. | 31.05.2007 |
| 10014903 | BAYO S.A.S. | EASA.A.S.03138 | | | CESSNA P210N | 53/2006 - IFR B-RNAV certification extension of the GNS530 already installed. | | Flight Manual Supplement 53/2006-AFMS Ed. 1 Rev. 1 dated 20 September 2007. | STC Bayo 36/2006 must be already installed.The aircraft must be equipped with HSI KI525A and S-TEC 55X autopilot. | 21.01.2008 |
| 10014920 | BAYO S.A.S. | EASA.A.S.03167 | | | CESSNA 177RG | STC 42/2007 - Modification of radio electric installation for B-RNAV certification with AP coupling. | | Airplane Flight Manual Supplement Ed1 Rev1, dated 20 September, 2007. | 1. Aircraft must be equipped with HSI KI525A and S-TEC 60-2 autopilot;2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification (...) | 08.02.2008 |
| 10015010 | BAYO S.A.S. | EASA.A.S.03312 | | | CESSNA 182J | 54/2006 - IFR B-RNAV certification extension of the GNS430 without Autopilot coupling. | | Flight Manual Supplement 54/2006-AFMS Ed. 1 Rev. 1, dated 23 January 2008. | STC Bayo 27/2006 must be already installed.The Aircraft must be equipped with HSI KI525A. | 24.01.2008 |
| 10015098 | BAYO S.A.S. | EASA.A.S.03439 | REV. 1 | | FAIRCHILD SWEARINGEN SA 226 T | Modification of radio electrical Installation:2 VHF 21A, 2VOR/ILS/MKR VIR 30A, 2 ATC TDR 90, 1 RMI 30 removal;2 GNS 430 W.2 MD 41, 2 A-30 ACK, 2 GTX 330D, 1 KA-33, 1 ST3400 Installation | | Airplane Flight Manual Supplements:- 64/2007-AFMS TAWS Ed. 1 Rev. 0 Feb. 2008- 64/2007-AFMS GNS 430 W/Limitation- Ed. 1 Rev. 0 Feb 7 2008. | The approval is limited to MSN T-276 F-GGLOGPS use is forbidden for SID/STAR/Approaches/GPS/Auto Pilot coupling is forbidden below 2000 ft. | 05.03.2008 |
| 10015138 | BAYO S.A.S. | EASA.A.S.03491 | | | FAIRCHILD SWEARINGEN SA 226 T | Modification of radio electrical installation: replacement radar indicator by MFD Garmin GMX200 and/or Concord Batteries. | | Airplane Flight Manual Supplement: 06/2008-AFMS GMX 200/Limitation- Ed. 1 Rev. 0 Mars 2008. | 1. The approval is limited to MSN T-276 F-GGLO;2. Use of GMX 200 limited to radar weather display function;3. Prior to installation of this modification the installer must determine that the interrelationship between this modification and an (...) | 08.04.2008 |
| 10015179 | BAYO S.A.S. | EASA.A.S.03557 | | | PIPER PA-25-235 | Modification of radio electrical installation.Installation of AG-NAV2 GPS spreading guidance system not to be used for navigation, in accordance with "Fiche d'approbation de conception d'une évolution No. 53/2007, rev.3 dated 12/12/08". | | None. | Limited to S/N 25-74-05726, 25-3109, 25-3704, 25-3984, 25-4382, 25-3292,25-2400, 25-4445. | 21.01.2009 |
| 10015220 | BAYO S.A.S. | EASA.A.S.03635 | | | BEECH 58 | Modification of radio electrical installation for B-RNAV certification with AP coupling | | Dossier de Définition STC 43/2007- DEF Révision 3 datée du 05/11/2008Supplément au Manuel de Vol Beech 58 pour #utilisation du GPS GNS 530 certifié en moyen supplémentaire de navigation IFR/VFR avec la capacité Basic RNAV couplé au pilote automatique (...) | The applicability of this STC is limited to the Beech 58 S/N TH-48N Navigation on SID and STARs prohibited in Terminal Area Utilisation of GPS prohibited for approaches Prior to installation of this modification the installer must determine that the in (...) | 14.11.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|-------------------|----------------|----------------------|--|--|--|--|--|---------------|
| 10017122 | BAYO S.A.S. | EASA.R.S.01209 | | | EUROCOPTER FRANCE AS 365N SERI | BAYO avionics modification no. 16/2005: installation of two NAV/COM/GPS GNS 430 for IFR B-RNAV operations, with their associated MD 41 annunciators, A-30 ACK altitude encoders and KA 33cooling system, a Garmin GTX 330 mode S transponder, two Bend (...) | | Definition and installation: BAYO definition file n° 16/2005-DEF, revision 0, dated December 19th, 2005, or later EASA approved revision. Operation: BAYO RFM supplement no. 16-2005-AFMS: Edition, Revision 0, dated December 19th, 2005, or later EASA app (...) | This STC can be installed only by BAYO technical personal in accordance with BAYO STC definition file n° 16/2005-DEF, revision 0, dated December 19th 2005 or later EASA approved revision | 19.01.2006 |
| 10032968 | BAYO S.A.S. | | | STC 2010-006 | HR 100/200, HR100/200 B HR 100/210, HR 100/210 D HR 100/285 TIARA HR100/250 TR, HR 100/285 C | Installation of single EFIS Aspen EFD1000 Pro for VFR navigation. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Bayo AFM supplement n° 2010-006, revision 3 dated 09/12/2010 Bayo Master list n° 2010-006-ML, revision 4 dated 09/12/2010 Bayo modification file n° 2010-006-DEF, revision 2 dated 30/09/2010 | Approved software version for ACU and PFD units is 1.1. Navigation coupling to automatic pilot is limited to heading mode. | 10.12.2010 |
| 10015726 | BAYSYS TECHNOLOGIES, LLC | EASA.IM.A.S.01741 | | FAA STC ST01939LA | AIRBUS A319-115 AIRBUS A319-133 | Direct TV - Installation of a tailwind 55 Satellite direct TV receive system | | Master Data List (MDL), ref: 8050-ED-1067, revision L, dated November 20, 2006 or later approved revisions. Airplane Flight Manual Supplement (AFM-S), ref: 8050-AFMS, revision IR, dated November 20, 2006 or later approved revisions | 1. Maximum Takeoff Weight and Maximum Landing Weight shall be reduced as follows: Takeoff 350 lbs (159 Kg). Landing 410 lbs (186 Kg). For airplane(s) modified by this STC, prior to the accumulation of 9,200 total flight cycles from the STC modification (...) | 23.11.2006 |
| 10013982 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.01014 | | | AIRBUS A320-200 | Reconfiguration of Passenger Cabin to 180 Economy Class Passengers | | -BBJ394 Issue 1 or later approved revision | | 17.01.2005 |
| 10013985 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.01017 | REV. 1 | | BOEING 737-700 BOEING 737-800 | Installation of 44 Passenger Seats, seats power supply and in seat power supply system on Boeing-BBJ128 Reynard J2000 double seats, safety equipment, carpet and new wardrobes. Installation of General Dynamics In Seat Power System. | | Modification Document BBJ385 Issue 4 or later approved revision | | 24.10.2005 |
| 10013986 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.01018 | | | | Floor Structure rework. | | BBJ Design Services Ltd Modification Summary Sheet BBJ365 dated May 18, 2005 or later EASA approved revisions. | 1. This STC is restricted to Boeing 767-329 Registration G-VKNG, Serial Number 23785.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with oth (...) | 08.06.2005 |
| 10013987 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.01019 | | | | Replacement of a Lufthansa Technik IFE System with an ITAN IFE System | | -BBJ383 Issue 2 dated 21 February 2005 or later approved revision-Flight Manual Supplement BBJFMS014 Issue I/R | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 17.03.2005 |
| 10014047 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.01129 | | | BOEING 757-200 | Enhanced Mode S Surveillance System completion and activation | | -Modification Summary Document BBJ342 Issue 1 21/09/04 or later EASA approved revision -AFM Supplement BBJFMS012 Revision IR or later EASA approved revision | | 01.04.2005 |
| 10014048 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.01130 | | | BOEING 767-300 | Enhanced Mode S Surveillance System completion and activation | | -Modification Summary Document BBJ343 Issue 1 21/09/05 or later EASA approved revision -AFM Supplement BBJFMS013 Revision IR or later EASA approved revision | | 01.04.2005 |
| 10014232 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.01395 | | | AIRBUS A321 | Installation of Koito 24B/C and 148Y seats for Dragonair | | BBJ Certification Plan BBJCP015 | | 11.11.2005 |
| 10014321 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.01573 | | | AIRBUS A330 | Installation of In Seat Power Supply Korean Air Airbus A330 | | BBJ Design Services Modification BBJ569 | | 25.01.2006 |
| 10014388 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.01674 | | | | Replacement of Aft G4B Galley. | | BBJ Design Services Modification BBJ632 | | 27.06.2006 |
| 10014520 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.02601 | REV. 1 | | BOEING 767-300ER | Installation of a MAS2000E in Flight Entertainment System on Austrian Airlines B767-300ER aircraft | | BBJ Design Services Modification Reference BBJ705N0280-D70 Instructions for Continued Airworthiness NO280-D71: Maintenance Manual Supplement | -- | 02.02.2007 |
| 10014567 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.02668 | | | BD-700-1A11 | Installation of an Electronic Flight Bag (EFB) on A/C S/N 9089 Modification Summary List: BBJ745 issue 1CCL: BBJCCL205 issue 1ADoC: Mod. Sum. Sh. & Certificate of Design iss. 25/10/2006 | | AFM-Supplement: BBJFMS017 Rev IR Installation Manual: 929-600040-00 Rev 4 | This approval of this installation does not include any approval of the suitability, integrity, or accuracy of the EFB hardware, software, data base, display, controls or lighting. The capability of the system to perform its intended function has n (...) | 03.11.2006 |
| 10014658 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.02815 | | | BOEING 747-400 | Interior changes required on Cathay Pacific Boeing 747-400 aircraft MSN 27132 for BE Aerospace Flight Structures. | | MOD BBJ 804 | | 10.05.2007 |
| 10014665 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.02822 | REV. 1 | | BOEING 757-200 SERIES | Cabin Reconfiguration on Boeing 757-230 aircraft | | BBJ Design Services Ltd Modification BBJ807. | -- | 02.03.2007 |
| 10014770 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.02962 | | | BOEING 747-400 | Interior Changes required on Cathay Pacific Boeing 747-400 for BE Aerospace Flight Structures. | | BBJ Design Services Modification reference BBJ865. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 02.07.2007 |
| 10014774 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.02966 | | | BOEING 767-300 | Installation of an in-seat power supply system and reconfiguration of business class seating. | | BBJ MOD No. BBJ840 Issue 2 or later. Maintenance Manual Supplement BBJ MS057. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 30.05.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|-------------------|----------------|-----------------------|-------------------------------|---|---|---|--|---------------|
| 10014784 | BBJ DESIGN SERVICES LIMITED | EASA.A.S.02984 | | | BOEING 757-200 | Installation of fixed Emergency Locator Transmitter | | Modification BBJ866, Technical Publication Supplements BB/JMS058. | N/A | 13.04.2007 |
| 10014250 | BEECHCRAFT BERLIN AVIATION GmbH | EASA.A.S.01472 | REV. 2 | | B200 | Installation of Cloud Research and Seeding System This EASA STC 10014250<->-> REV. 2 reflects the transfer of STCEASA.A.S.01472 Revision 1 from Scandinavian Avionics Design GmbH to Beechcraft Berlin Aviation GmbH | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Airplane Flight Manual Supplement Doc 2023287-AM-01, Issue 5 dated August 2005. Continued Airworthiness Doc. 2023287-CA-01 Engineering Order Doc 2023287-EO-01 or later revisions of the above listed documents approved by EASA | None | 27.09.2010 |
| 10042830 | BEECHCRAFT BERLIN AVIATION GmbH | | REV. 1 | | C90GTI | Research Mission Applications Change to STC to reflect allowed FK1 for Configuration A (Hawkeye and CAPS Probe installed) | The Certification Basis of the original product and the following additional / alternative airworthiness requirements are applicable to this change: CS-23, Initial Issue. The certified noise and/or emissions levels of the original product are unchanged (...) | EASA approved Master Data List, Doc. No. D11-B13-01, Issue 01, dated 20-Jun-2013<->->EASA approved Airplane Flight Manual Supplement, Doc. No. D30-B13-01, Issue 01, dated 11-Jun-2013<->->Maintenance Manual Supplement, Doc. No. D31-B12-01, Issue 01 (...) | For mission equipment installed under the wing speeds are limited to V _{NO} = 150 KIAS, and V _{NE} = 189 KIAS. For limitation on lights into areas with known icing conditions, and further limitation refer to AFMS, Doc. No. D30-B13-01, Section II Installation (...) | 12.07.2013 |
| 10043630 | BEECHCRAFT BERLIN AVIATION GmbH | | | | B300 | Installation of Flight Inspection System UNIFIS 3000. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | EASA approved Master Document List, Doc. No. D11-B11-02, Issue 02, dated 31-Jan-2013<->->EASA approved Airplane Flight Manual Supplement, Doc. No. D30-B11-02, Issue 05, dated 24-Jan-2013<->-> or later revisions of the above listed documents approved (...) | For limitations see AFMS, Doc. No. D30-B11-02. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse (...) | 13.02.2013 |
| 10016501 | BEECHCRAFT CORPORATION | EASA.IM.A.S.02863 | REV. 3 | FAA STC SA10747SC | C90, C90A, C90GT, C90GTI, E90 | Increase Operating Gross Weight Approval (as defined in the attached table) | FAR 23 | - Increase in gross weight installation in accordance with Cen Tex Aerospace Incorporated (Inc.) Drawing No 030-0000-0000, Rev G or later approved revisions. - Operation in accordance with Cen Tex Aerospace Incorporated (Inc.) Aircraft Flight Manual Suppl (...) | This modification is only applicable to aircraft equipped with the Collins Proline 21 avionics package. Refer to aircraft defined in Hawker/Beechcraft major change SC0353A. | 23.03.2011 |
| 10016625 | BEECHCRAFT CORPORATION | EASA.IM.A.S.03058 | REV. 1 | FAA STC SA03622AT | BEECH B300, BEECH B300C | Installation of an Air Conditioning Evaporator Blower Motor RF Noise Filter into a Hawker Beechcraft Model B300 / B300C aircraft . | | In accordance with Master Data List Number HBS 003-2151-101-00 Revision 1.0, dated 09/16/08 or later approved revision. | This modification is only applicable to aircraft equipped with the Collins Proline 21 avionics package. Refer to aircraft defined in Hawker/Beechcraft major change SC0353A. | 23.03.2011 |
| 10027658 | BEECHCRAFT CORPORATION | | REV. 1 | FAA STC SA02253LA | B300/B300C | Installation of Belly Radome. | FAR 23 The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Installation in accordance with Master Data List 300E36871 Revision 08, dated September 14, 2009, or later approved revisions. - Operation in accordance with Airplane Flight Manual Supplement, 130-590031-327, September 2009, for 16,500 lb MTOW Config (...) | none | 05.10.2011 |
| 10029215 | BEECHCRAFT CORPORATION | | | FAA STC SA01502W1-D | B300 SERIES | STC SA01502W1-D issued to allow modification of the FGC-3000 units as solution to eliminate nuisance pitch trim wheel movement. | FAR 23. The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Instructions Replacement of FGC-3000 in a Hawker Beechcraft Corporation King Air B300/B300C, P/N RCA-2463, dated January 28, 2008. Instructions for Initial and Continued Airworthiness, P/NAMI-STC-B2B3P21, Rev 9 dated September 18, 2009 or I (...) | None. | 12.03.2010 |
| 10029220 | BEECHCRAFT CORPORATION | | | FAA STC ST03749AT-D | GIV-X | Installation of cabin interior and Airplane Completion for Serial 4175. | JAR 25. The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Install in accordance with Master Index List GC414914175, Revision H, dated January 27, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | This approval is limited to only the installation made in Gulfstream GIV-X, Serial No 4175. The following STCs are a required part of this STC: ST022290H-D, ST02979AT-D, ST02980AT-D, ST02987AT-D, ST03047AT-D, ST03290AT-D, ST03410AT-D, ST03532AT-D, (...) | 12.03.2010 |
| 10029784 | BEECHCRAFT CORPORATION | | | FAA STC REF SA02277LA | B300 KING AIR | Installation of High Definition Audio Video (HDAV-3002) LRU Unit as an interface with the Rockwell Collins Venue Cabin Management System. | FAR 23 The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Install in accordance with Rockwell Collins System Drawing List ref441-0130, Rev B, dated December 15, 2009. Continued Airworthiness Document, General Aviation - Rockwell Collins Cabin Systems, P/N 442-1306-001, Rev New, dated 11/11/09 or later revision (...) | Only applicable to King Air Model B300 aircraft equipped with the Rockwell Collins Venue Cabin Management System. | 20.04.2010 |
| 10030002 | BEECHCRAFT CORPORATION | | | FAA STC SA10969SC | KING AIR C90GTI | Installation of an upgraded Flight Management System (Version 4.0). This upgrade will offer a standard Flight Management Computer and will include an option for Search and Rescue Mission Profiles. | FAR 23 The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with in accordance with master drawing list 337-00-0001, revision IR, dated January 27, 2010, or later approved revisions. Operation in accordance with Airplane Flight Manual Supplement for Collins FMS-3000 LPV Upgrade, P/N 33 (...) | None. | 10.05.2010 |
| 10030003 | BEECHCRAFT CORPORATION | | | FAA STC SA10970SC | KING AIR B200 AND B300 | Installation of an upgraded Flight Management System (Version 4.0) . This upgrade will offer a standard Flight Management Computer and will include an option for Search and Rescue Mission Profiles. | FAR 23 The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Installation in accordance with in accordance with master drawing list 337-00-0001, revision IR, dated January 27, 2010, or later approved revisions. Operation in accordance with Airplane Flight Manual Supplement for Collins FMS-3000 LPV Upgrade, P/N 337 (...) | None. | 10.05.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|------------------------------|----------------|----------------------|-------------------------------------|--|---|--|--|---------------|
| 10035154 | BEECHCRAFT CORPORATION | | REV. 1 | FAA STC SA00526DE | 390 (PREMIER I) 390 (PREMIER IA) | Installation of an AirCell telephone system. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Master Data List, Document No.GEN-030317-MDL, Revision B dated September 5, 2003. Maintain in accordance with Document No GEN-030317-018 Rev. A, dated September 5, 2003. Aircraft to be operated in accordance with Flight (...) | Serial Number RB-73 Only Hawker Beechcraft Ltd. Minor Modification SC0392A must be embodied. | 15.11.2012 |
| 10036742 | BEECHCRAFT CORPORATION | | | FAA STC SA02331LA | B300 & B300C | Installation and maintenance of Switches and wiring for an Electro-Optical/Infra-Red Lift and Turret System. | FAR 23 The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Master Data List 300E404584, Rev 3, dated November 9, 2010. Operation in accordance with Airplane Flight Manual Supplement for EO/IR Lift Electrical System, PN 130-590031-403, dated, December 3, 2010. or later revisions (...) | none | 05.10.2011 |
| 10017062 | BELL HELICOPTER TEXTRON CANADA, Ltd | EASA.IM.R.S.01497 | | TCCA STC NO. SH05-2 | BELL 212 BELL 412 BELL 412 EP | TCCA SH05-2 #Cold Weather Use of Kerosene Fuels#. | | 1. Model 212: BHT-212-FMS-31 Issue 0, dated 24 January 2005.2. Model 412: BHT-412-FMS-23.1/-23.2/-23.3 Issue 0, dated 24 January 2005.3. Model 412EP: BHT-412-FMS-23.4 Issue 0, dated 24 January 2005. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 24.11.2008 |
| 10016755 | BELL HELICOPTER TEXTRON, Inc. | EASA.IM.R.S.01084 | | SR09397RC | BELL 206A BELL 206B | STC ST09397RC Increase in takeoff weight from 1.451 Kg (3200lbs) to 1.520 Kg (3350 lbs when modified in accordance with Bell Model 206 Series Service Instructions BHT-206-SI-129 dated July 12, 2004, or later FAA approved revision. | | Rotorcraft Flight Manual Supplement for Increased Internal Gross Weight to 3350 lbs., BHT-206B3-FMS-37, Bell Helicopter Textron Inc., Original Revision, dated July 12, 2004. Instructions for Continued Airworthiness (ICA) for Bell Model 206B Series heli (...) | Aircraft effectivity is as follows: # This STC is limited to 206A or 206B aircraft that are qualified for and configured with the 65 inch diameter tail rotor. # 206A S/N 4 thru 660, and 672 thru 715 if modified to a Model 206B by incorporation (...) | 01.02.2006 |
| 10017021 | BELL HELICOPTER TEXTRON, Inc. | EASA.IM.R.S.01439 | | FAA STC SH8713SW | BELL 230 | Replacement of the cabin door with right hand sliding door. This STC has been classified as #BASIC# by the FAA and has been approved under the PTVF. | | 230-530-102 Revision A dated 28 April 1994 or later approved revision # Sliding Door Installation R.H.; 230-530-116 Revision IR dated 1 August 1994 or later approved revision # Sliding Door Installation L.H.; BHT-230-FMS-24 Reissue 0 dated 29 September (...) | | 08.05.2008 |
| 10037652 | BELL HELICOPTER TEXTRON, Inc. | | | FAA STC SR03923AT | BELL 407 | Installation of Aeronautical Accessories Lighting and Vision Accessories for Bell Helicopter Textron 407 Model Helicopters. Any of the following kits can be installed individually or in any combination: - Preflight Lighting Kit per installation instruction (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The following FAR 27 paragraphs: §27.301 Amdt 27-0527, §303 Amdt 27-0527, §305 Amdt 27-0527 (...) | -RFM Supplement, Aeronautical Accessories documents AA-10062 and AA-11004 for Bell 407 series Aircraft, dated June 29, 2011 or later revisions of the above listed documents approved by EASA-MDL A-2009-095-1, Rev C, dated June 27, 2011-Drawing List AA-1 (...) | None | 13.12.2011 |
| 10038134 | BELL HELICOPTER TEXTRON, Inc. | LBA STC RC1132 / 98/5476/MAE | REV. 1 | FAA STC SH3311SO | BELL 212, 412, 412EP | Replacement Windows | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AA92083 revision M dated 20 September 2010 - Drawing List and Installation Instructions. | None | 26.01.2012 |
| 10038194 | BELL HELICOPTER TEXTRON, Inc. | | | FAA STC SR02808NY | 429 | Installation of wedge Windows with Sliders into L/H and R/H Passenger Doors in accordance with Aeronautical Accessories Inc. Installation Instructions Report Number AA-09064, Revision A, 09. (FAA STC SR02808NY) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Aeronautical Accessories Inc. Installation Instructions Report Number AA-09064, Revision A or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) ICA Rep (...) | None | 26.01.2012 |
| 10038231 | BELL HELICOPTER TEXTRON, Inc. | | | FAA STC SR00707DE | 429 | Automatic Door Openers for Crew and Passenger Doors and Pilot and Co-Pilot Floor Protectors | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AA-09025 revision B dated 6 November 2009 - Drawing List for Door Openers, Floor Protectors and Paint Protector; AA-09045 revision B dated 7 July 2010 - Installation Instructions, Crew Floor Protector Kit; AA-09070 revision B dated 30 October 2009 - Cert (...) | None | 30.01.2012 |
| 10038305 | BELL HELICOPTER TEXTRON, Inc. | | | FAA STC SR09534RC-D | BELL 429 | Installation of Cargo Mirror | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AA-09030 revision Original dated 8 April 2010 - Rotorcraft Flight Manual Supplement or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) AA-09029 rev (...) | None | 06.02.2012 |
| 10039466 | BELL HELICOPTER TEXTRON, Inc. | | | FAA SR02810NY | 429 | Installation of Hardpoint and Spotter Fixtures in accordance with Aeronautical Accessories Inc. Installation Drawing List for Hardpoint Fixture Kit, Ceiling Spotter Fixture Kit, Floor Spotter Fixture Kit, Report Number AA-09043, Rev C, dated June 29, 2010. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Aeronautical Accessories Inc. Installation Drawing List for Hardpoint Fixture Kit, Ceiling Spotter Fixture Kit, Floor Spotter Fixture Kit, Report Number AA-09043, Rev C, dated June 29, 2010; Lawson Aerospace LLC Instructions for Continued Airworth (...) | None | 07.05.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|--------------------|----------------|----------------------|-------------------------------|--|--|--|---|---------------|
| 10041096 | BELL HELICOPTER TEXTRON, Inc. | | | FAA STC SR09537RC-D | 429 | Installation of High Frequency (HF) Antenna Kit (P/N 429-007-001) for Bell Helicopter Textron Model 429 Helicopters. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Rotorcraft Flight Manual Supplement Report No. AA-11021 Rev. NR, dated June 22, 2011, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) - Bell Helicopter (...) | Only the installation of the HF Antenna Kit is included in this STC. The antenna is capable of interfacing with many different types of HF radios; therefore, the compatibility of an HF radio installation with the HF Antenna Kit is the responsibility of (...) | 22.08.2012 |
| 10041097 | BELL HELICOPTER TEXTRON, Inc. | | | FAA STC SR00739DE | 429 | Installation of Sliding Door Push-Out Windows Kit (P/N 429-563-001/2/3/4/5/6/7/8/9) for Bell Helicopter Textron Model 429 Helicopters. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Drawing List Laz-Tec, LLC Report No. AA-11062 Rev. B, dated November 23, 2011, FAA Approved December 19, 2011, -Instructions for Continued Airworthiness Laz-Tec, LLC Document No. AA-11065 Rev. A dated December 02, 2011, FAA accepted December 06, 2011. | None | 22.08.2012 |
| 10041124 | BELL HELICOPTER TEXTRON, Inc. | | | FAA SR00740DE | BELL 429 | Installation of Aft Cabin Floor Protectors Kit (P/N 429-457-001) for Bell Helicopter Textron Model 429 Helicopters. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Drawing List Laz-Tec Report No. AA-11059 Rev. B dated December 07, 2011, FAA Approved December 16, 2011, -Instructions for Continued Airworthiness Laz-Tec, LLC Document No. AA-11061 Rev. A, dated December 02, 2011, FAA accepted December 06, 2011. | None | 22.08.2012 |
| 10041125 | BELL HELICOPTER TEXTRON, Inc. | | | FAA SR00741DE | BELL 429 | Removable Crew Door Installation Kit (P/N 429-650-001/3/4) for Bell Helicopter Textron Model 429 Helicopters. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Flight Manual Supplement, Document No. AA-11070, Revision NR, dated January 19, 2012, FAA Approved January 19, 2012, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions (...)) | None | 22.08.2012 |
| 10041215 | BELL HELICOPTER TEXTRON, Inc. | | | FAA STC SR09541RC | 407 | Passenger Sliding Doors Kit P/N 407-270-003 for Bell Helicopter Textron Model 407 Helicopters. Reference FAA STC SR0941RC. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Rotorcraft Flight Manual Supplement for Passenger Sliding Doors Kit P/N 407-270-003, Report No. AA-10101 Rev. NR, dated April 10, 2012, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or (...)) | 1. The Passenger Sliding Doors Kit P/N 407-270-003 modifies the hinge passenger doors for the Bell Helicopter Textron Canada Limited (BHTCL) Model 407 Helicopters. A Personnel attachment Fixture intended to the passenger compartment is not intended to (...) | 30.08.2012 |
| 10041217 | BELL HELICOPTER TEXTRON, Inc. | | | FAA STC SR09540RC | 407 | Crew Area Hardpoints Kit BHTI P/N 407-240-001 Internal Hardpoint Kit BHTI P/N 407-638-001 External Hardpoint Kit BHTI P/N 407-230-001 and P/N 407-230-100 Pylon Hardpoint Kit BHTI P/N 407-854-001/2/3 Reference FAA STC SR09540RC | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Rotorcraft Flight Manual Supplement for External Hardpoint Kit P/N 407-230-001 and P/N 407-230-100, Report No. AA-10114 Rev. NR, dated April 10, 2012, -Rotorcraft Flight Manual Supplement for Pylon Hardpoints Kit P/N 407-854-001/2/3 Report No. AA-1012 (...) | 1. High skid landing gear is required for BHTI P/N 407-638-001 External Hardpoint Kit installation. 2. Installation of BHTI P/N 407-270-003 Passenger Sliding Doors Kit (separate STC) is a co-requisite for BHTI P/N 407-854-001/2/3 Pylon Hardpoints Kit ins (...) | 30.08.2012 |
| 10041283 | BELL HELICOPTER TEXTRON, Inc. | | | FAA STC SR02405LA | 407 | Survivor Window Raft Kit P/N 407-567-002 for Bell Helicopter Textron Model 407 Helicopters. Reference FAA STC SR02405LA. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Rotorcraft Flight Manual Supplement to the Bell 407 Helicopter Flight Manual when equipped with the Survivor Window Raft Kit. Document No. AA-10078 Rev. NR, dated December 6, 2011, or later revisions of the above listed documents approved by EASA in accordance with (...) | None | 04.09.2012 |
| 10044970 | BELL HELICOPTER TEXTRON, Inc. | | | FAA STC SR09598RC | 206L, 206L-1, 206L-3, 206L-4 | Installation of a Radar Altimeter Antenna Mount Kit. This STC makes provisions for the Radar Altimeter Antennas installation in accordance with the Master Drawing List AA-12015. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Master Drawing List AA-12015 Rev. A, dated September 17, 2012, -Instructions for Continued Airworthiness AA-12016 Rev., dated August 14, 2012, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/ (...) | Only the installation of the Mount is approved. Installation of equipment for which these provisions are intended is not approved. It is the responsibility of the Radar Altimeter System installer to obtain separate approval for the Radar Altimeter System (...) | 23.05.2013 |
| 10044994 | BELL HELICOPTER TEXTRON, Inc. | | | FAA STC SR01952AT | BELL 412 | Cargo Door Retention Kit | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -RFM Supplement Report AA-99006, dated 19 July, 1999, -Aeronautical Accessories Inc. Drawing List Report Number AA-98170 dated February 5, 1999, -Aeronautical Accessories Inc. Installation instructions - Report AA-98171, Dated January 19, 1999, -Aer (...) | S/N 36191 and subsequent Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product | 24.05.2013 |
| 10015578 | BENZ AIRBORNE SYSTEMS | EASA, IM.A.S.01489 | | FAA STC SA10365SC | PIPER PA-31T PIPER PA-31T1 | FAA STC SA10365SC: Installation of a Benz Airborn System Model 800-0021 torque pressure system. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Benz Engineering Order (Instruction for Installation) SB079-4021-00, Rev. B, dated 19 January 2005. | N/A | 19.03.2007 |
| 10030405 | BENZ AIRBORNE SYSTEMS | | | FAA STC SA01989CH | DHC-6-100, -200, -300 | Installation of Torque Pressure System on DHC-6 in acc. to FAA STCSA01989CH | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Armstrong Aerospace Master Data List No. DC104-3021-00, Rev. B, dated July 18, 2005, Benz Airborne Systems, Engine Torque Pressure Indicating System DHC-6-300, Aircraft Maintenance Manual Supplement, Rev A, March 10, 2004, or later revisions of the above (...) | For models DHC-6-100 and DHC-6-200, this installation is only applicable when the airplane has been modified with PT6A-27 engines. | 15.06.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|----------------|----------------|----------------------|---|--|--|--|--|---------------|
| 10030408 | BENZ AIRBORNE SYSTEMS | | | FAA STC SA01970CH | DHC-6-100, -200, -300 | Installation of Engine Oil Pressure System on DHC-6 in acc.to FAA STCSA01970CH | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Armstrong Aerospace Master Data List No. DC104-3020-00, Rev. B, dated July 18, 2005; Benz Airborne Systems, Engine Oil Pressure Indicating System DHC-6-300, Aircraft Maintenance Manual Supplement, Rev A, March 10, 2004 or later revisions of the above list (...) | For models DHC-6-100 and DHC-6-200, this installation is only applicable when the airplane has been modified with PT6A-27 engines. | 15.06.2010 |
| 10029629 | BERINGER AERO | | | | PC-6/A2-H2, PC-6/B1-H2 PC-6/B2-H2, PC-6/B2-H4 | BRGAP-STC-002 - New wheels and brake system. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | None. | None. | 08.04.2010 |
| 10041707 | BERINGER AERO | | | | DISCUS-2A, DISCUS A, VENTUS-2A, VENTUS A. | BRGAP-STC-007(0) - Replacement of wheel and brake system | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual supplement FM-STC-007(00)-12.09.2012 or later revisions of the above listed documents approved by EASA. | None | 10.10.2012 |
| 10042599 | BERINGER AERO | | | | 1- DHC-6 SERIES 1 1- DHC-6 SERIES 100 1- DHC-6 SERIES 110 1- DHC-6 SERIES 200 1- DHC-6 SERIES 210 1- DHC-6 SERIES 300 1- DHC-6 SERIES 310 1- DHC-6 SERIES 320 2- DHC-6 SERIES 400 | Replacement of nose wheel BF Goodrich 8.90-12.5 P/N 3-1197 by Beringer wheel assembly P/N: RA-008. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Configuration Dossier DM-STC-001, Rev. 2, dated December 7, 2012; Installation and Maintenance Manual MM-STC-008, Rev. 1, dated December 4, 2012; Classification Sheet FCM-STC-008, Rev. 0, dated October 30, 2012; Test Report RET-STC-008, Rev. 1, dated Dec (...) | None. | 10.12.2012 |
| 10043080 | BERINGER AERO | EASA.A.S.03857 | | | APEX DR 400 TOUS MODELES R1000, R2000, R3000, HR100, | New Design of brake Disks | | PET-STC-001 Certification Program PET-STC-001 Test Report Drawing AV-ROB-001-01 | none | 05.05.2009 |
| 10043346 | BERINGER AERO | | REV. 1 | | SR20, SR22, SR22T | STC-004(1) dated 10 January 2013 - Installation of RF-006(A) main wheelwell and RA-002(A) nose wheel well with EA-003.4N(A) brake system on Cirrus Design Corp. SR20, SR22, SR22T. Revision 1: Upgrade of brake assembly EA-003.4N to the (B) revision and (>, (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 23 amdt 3. The requirements for environmental protection and the associated certific (...) | Airworthiness limitation information, §3 - Life limited parts, of the MM-STC-004(1) BRGAP-STC-004(02) dated 24/07/2013; MM-STC-004 rev 2 dated 16.07.2013; MM-STC-004 rev 1 dated 16.07.2013; NP-STC-004 rev 1 dated 16.07.2013 or later revisions of the above listed documents approved by EASA. | The approval holder shall fulfill the obligations of Part 21, Paragraph 21.A.109. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or (...) | 05.09.2013 |
| 10043876 | BERINGER AERO | | REV. 1 | | DA40NG | STC-009(1) dated 04 Feb 2013 - Installation of RF-006(A) main wheel and RA-005(A) nose wheel with EA-004N(A) brake system on DA40 NG. This Rev. 1 was issued to correct typing error in the description of Design Change. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 23 Amendment 1. The certificated noise and/or emissions level of the original product (...) | Airworthiness limitation information, §3 - life limited parts, of the MM-STC-009; Dossier de modification DM-STC-009(1) dated 04 February 2013; Maintenance and Overhaul Manual MM-STC-009(0) dated 21 January 2013; Part List NP-STC-009(0) dated 04 Februar (...) | This STC is valid only on versions with OAM 40-334 "landing gear with large tyres". Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change a (...) | 30.08.2013 |
| 10045516 | BERINGER AERO | | REV. 1 | | SR20, SR22, SR22T | STC-010(0) dated 16 May 2013 - Installation of RF-006(A) main wheel and RA-002(A) nose wheel with EA-003.3N brake system on Cirrus SR20, SR22 and SR22T | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Airworthiness Limitations Information, §3 - Life limited parts, of the MM-STC-010(0) Dossier de modification DM-STC-010(0) dated 16 May 2013; Maintenance and Overhaul manual MM-STC-010(0) dated 13 May 2013; Part List NP-STC-010(0) dated 13 May 2013 or late (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce adverse effect upon the airworthiness of the product. | 08.07.2013 |
| 10036735 | BHE & ASSOCIATES, LTD. | | | | B200, B200C, B200CGT, B200GT B300, B300C | Installation of upgraded hardware to the Rockwell Collins Pro Line 21 FMS-3000 to provide Localiser Performance with Vertical Guidance (LPV). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Master Drawing List 324-00-0001, revision A, dated January 12, 2010; Operation in accordance with FAA approved Aircraft Flight Manual Supplement 324-00-0014, revision IR, dated January 12, 2010, and EASA Approved Aircraft (...) | This installation is limited to aircraft serials listed in BHE and Associates document 324-00-0002, Rev IR, dated November 24, 2009, or later approved revisions. | 05.10.2011 |
| 10045378 | BHE & ASSOCIATES, LTD. | | | FAA STC ST11041SC | MF50, F50EX | Installation of upgraded hardware to replace Rockwell Collins Pro Line 4 displays with Rockwell Collins Pro Line 21 displays, add an Integrated Flight Information System (IFIS) and enable localizer performance with vertical guidance (LPV) capability. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): High Intensity Radiated Fields (HIRF) Protection as per CRI F-5107-02 closed at issue 01. The requirement (...) | Master Drawing List 360-00-0001 Rev. Q dated June 07, 2013; Airplane Flight Manual Supplement 360-00-0056 Rev. D dated June 13 (> <-) 2013; Airplane Flight Manual Supplement for EASA registered airplanes 360-00-0073 Rev. IR dated June 13, 2013 or later re (...) | This installation requires the installation of STC ST01433 W-D (EASA validated STC EASA.M.A.S.02889) in order to provide global positioning system data enhanced with the wide area augmentation system. Prior to installation of this design change it must (...) | 20.06.2013 |
| 10014464 | BINDER MOTORENBAU GmbH | EASA.A.S.02525 | | | ALEXANDER SCHLEICHER ASH 25 E | Modification of outer wing panel with winglets and 27 m wingspan | | ASH 25 E Technische Mitteilung Nr. 18/B1 | | 02.03.2006 |
| 10014471 | BINDER MOTORENBAU GmbH | EASA.A.S.02532 | | | ALEXANDER SCHLEICHER ASW 22 BL | Binder Motorenbau Technische Mitteilung ASW 22 BLE B2 Changes in engine installation - Different type of spindle drive - Installation of a primer ball-CFRP gear mount | | Binder Motorenbau Technische Mitteilung ASW 22 BLE B2 Drawings and Manual pages according to Tech. Note. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 12.04.2006 |
| 10014490 | BINDER MOTORENBAU GmbH | EASA.A.S.02552 | | | ALEXANDER SCHLEICHER ASH 25 | Modification of outer wing panel with winglets and 27 m wingspan | | ASH 25 Technische Mitteilung Nr. 17/B1 | | 03.02.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|-------------------------|--|---|---|---|--|---------------|
| 10029418 | BINDER MOTORENBAU GmbH | P-EASA.A.S.03569 | | | NIMBUS-3DM | Nimbus-3DM - Umbau von Rotax535C auf Solo 2625/2 | JAR 22 | Umbau Rotax 535C auf Solo2526/2 - Nimbus-3DM - W. Binder vom 12.02.2010Zusatz zum Wartungshandbuch für den Motorsegler Nimbus-3DM Umbau vonRotax535C auf Solo 2625/2 - 24.04.2009Zusatz zum Flughandbuch für den Motorsegler Nimbus-3DM Umbau vonRotax535C auf Solo 2625/2 - 24.04.2009 | None. | 24.03.2010 |
| 10034252 | BINDER MOTORENBAU GmbH | | | | ASH 25 E | Conversion from hand starter to electric starter.Umbau Handstarter auf Elektrostarter. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated-certificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | STC "ASH 25 E E-Starter", issue 27.01.2011 included drawings no.1-9Ergänzende Musterprüfung "ASH 25 E E-Starter", Ausgabe 27.01.2011einschliesslich Zeichnungen 1-9Addition of the Flight Manual, on STC "Retrofit Electric starter for thepowered sailpl (...) | None. | 21.03.2011 |
| 10036110 | BINDER MOTORENBAU GmbH | | | | ETA | | Design:Non-significant changeThe Certification Basis of theoriginal product remains applicable tothis change.Environmental Protection:The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Binder Motorenbau STC "eta Umrüstung SKB" 12.02.2011Beschreibung der Umrüstung und Zeichnung eta 2.023aErgänzung des Flughandbuches betreffend STC "Umrüstung SKB" für denMotorsegler eta 12.02.2011. | None. | 11.08.2011 |
| 10042442 | BINDER MOTORENBAU GmbH | | | | DG-500 M | DG500M - Umbau von Rotax535C auf Solo 2625/2 | The Certification Basis for the original product remains applicable tothis certificate/ approval.This certificate/ approval involves a change to the certificated noiseand/ or emissions levels. | DG500M - B1 - Umbau von Rotax 535C auf Solo2526/2;Zeichnungen lt. Zeichnungsliste DG-500 M (Stand 20.03.2012);Zusatz zum Wartungshandbuch für den Motorsegler DG-500 M STC DG500M-B1Umbau von Rotax535C auf Solo 2625/2 - 20.03.2012;Zusatz zumFlughandbu (...) | None. | 30.11.2012 |
| 10028420 | BIZJET INTERNATIONAL | | REV. 1 | FAA STC ST10964SC | A318-112 | Installation of provisions for an Executive Power System | Certification Basis as per EASA TCDS A.064The CertificationBasis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval- CRI H-01/Special Condition H-01 "Enhanced Ai (...) | - FAA Supplemental Type Certificate ST10964SC- Master Data List, ref. doc. 1110000100, Revision D, dated 17th August2010- Airplane Flight Manual Supplement, ref. doc. 1112021090, Revision IR,dated 18th December 2009- ICA Source Document, ref. doc. 1 (...) | Not applicable | 14.10.2010 |
| 10028421 | BIZJET INTERNATIONAL | | REV. 1 | FAA STC ST10963SC | A318-112 | Installation of a Miltope Wireless Access Point | Certification Basis as per EASA TCDS A.064The CertificationBasis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval- CRI H-01/Special Condition H-01 "Enhanced Ai (...) | - FAA Supplemental Type Certificate ST10963SC- Master Data List, ref. doc. 1110000200, Revision L, dated 13th July2010-Airplane Flight Manual Supplement, ref. doc. 1111021090, Revision IR,dated 18th December 2009- ICA Source Document, ref. doc. 1110 (...) | Not applicable | 14.10.2010 |
| 10028425 | BIZJET INTERNATIONAL | | REV. 10 | FAA STC ST10958SC | A318-112 A319-133 / -115 | Major Change to EASA STC 10028425 Rev 9 introducing changes in the Cabin/VP Interior Installation (add the -009 configuration)-...Installation of an Executive Style interior on A318-112,-Introduction of the Top Level Certification Summary | The Certification Basis for the original product as amendedby thefollowing additional or alternative airworthiness requirements:Special Condition(s) "...,CRI H-01 "Enhanced Airworthiness Programme for Aeroplane Systems -ICA on EWIS"-,...CRID-5302-007 (...) | - FAA Supplemental Type Certificate ST10958SC amended August 05, 2013- "Master Data List, ref. doc. 1110000010, Revision W, dated August 23,2013,-Airplane Flight Manual Supplement, ref. doc. 1350021091, Rev IR,dated June 14th, 2013- "Electrical WI (...) | Prior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noadverse effect upon the airworthiness ofthe product. | 20.09.2013 |
| 10042292 | BIZJET INTERNATIONAL SALES AND SUPP | | | FAA STC ST09033SC | MYSTERE FALCON 50 | Installation of a quick drain kit on Environmental Control UnitValidation of FAA STC ST09033SC | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associated-certificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Bizjet Master Drawing List N° BJ21-215001, Rev. A dated 06 Feb. 1995or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF(or subsequent revisions ofthis decision). | None | 21.11.2012 |
| 10016472 | BJAC BUSINESS JET AIRCRAFT | EASA.IM.A.S.02818 | | TC STC SA07-129 ISSUE 1 | BOMBARDIER BD-700-1A10 BOMBARDIER BD-700-1A11 | Goodrich Class III Electronic Flight Bag. | | TCCA STC No. SA07-129 Issue 1 dated 20 November 2007.TCCA SOC 5010-O-08-0020 dated 08 October 2008.Avionics Design Services Ltd. Master Drawing List MDL06079 Revision N/C, dated 2November 2007, or later approved revision.BJAC Services Flight Manual (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 14.10.2008 |
| 10015400 | BLACKHAWK MODIFICATIONS Inc. | EASA.IM.A.S.01110 | | FAA STC SA10341SC | BEECH C90 BEECH C90A BEECH E90 | Installation of Pratt & Whitney Canada PT6A-135A (de-rated to 550 HP) engines and McCauley 4HFR34C762/94LMA-4 or 4HFR34C768/94LMA-4 propellers in accordance with Blackhawk Modifications Inc Drawing No 19005-01 Rev IR, dated 15 March 2004, "Installation Instructions" or later approved revision | | AFMS No 19005 Rev IR, dated 24 November 2004 or later approved revisionNote: approved noise level 74.4 dB (A) at 10100 lbs | 1. AFMS No 19005, Rev IR 24 November 2004 or later approved revision is required for the Raytheon C90, C90A or E90 airplane with Pratt & Whitney Canada PT6A-135A engines. Note thatthe Horsepower is limited to 550 at 1900 RPM.2. Each enginmount must be removed from the airplane, cleaned and non-des | 29.03.2006 |
| 10015414 | BLACKHAWK MODIFICATIONS Inc. | EASA.IM.A.S.01134 | REV. 1 | FAA STC SA10341SC | BEECH C90 BEECH C90A BEECH E90 | Installation of Pratt & Whitney PT6A-135A engine with:1. Hartzell propellers HC-D4N-3C/D9290(S,K) with a diameter of 93inches (4 blades) in accordance with Blackhawk Modification, Inc.Drawing No 19003-01, Rev B, dated 1st April 2004, "Installation I (...) | | AFMS No 19003-00, Rev A, dated 7th October 2004 or later approved revision | AFMS No 19003-00, Rev A, dated 7th october 2004 or later approved revision is required for the Raytheon C90, C90A or E90 airplane with Pratt & Whitney Canada PT6A-135A engines. Note that the shaft horsepower is limited to 550 at 1900 RPMEach engine m (...) | 20.04.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|-------------------|----------------|-------------------------------|--|--|---|--|---|---------------|
| 10016078 | BLACKHAWK MODIFICATIONS Inc. | EASA.IM.A.S.02262 | | FAA STC SA10567SC | | Installation of Pratt and Whitney PT6A-135A engines in place of PT6A-135 engines. | | Installation must be in accordance with Master Drawing List No. 19015-00-00 Rev. IR dated 26-06-06 or later Approved Revisions. Operation must be in accordance with Flight Manual Supplement Document No. AFMS19015 Rev. IR dated 18-10-06 or later approv (...) | 1. Installation limited to Serial Number LA-2 thru LA-204, except for LA-202.2. Compatibility of this modification with other previously approved modifications must be determined by the installer.3. This STC is approved only for the product (...) | 23.01.2007 |
| 10016079 | BLACKHAWK MODIFICATIONS Inc. | EASA.IM.A.S.02263 | REV. 1 | FAA STC SA00433AT | BEECH 200C BEECH 200CT BEECH 200T | Installation of Pratt and Whitney PT6A-42 engines in place of PT6A-41 engines. This project is to amend EASA TSC No. EASA.IM.A.S. 02263 using the data from the amended FAA-STC SA00433AT dated 22. August 2007 to include Beechcraft Super King Air 200C, 200T and 200CT. | | Installation of Pratt & Whitney PT6A-42 engines with Propellers: model HC-B3TN-3(G,N)/T10178(B, NB, K, NK) or Hartzell HC-D4N-3A/D9383K installed in accordance with modifications drawing No. A794-0001, Rev. B, dated 17. July 2007 or later approved r (...) | | 01.04.2008 |
| 10016156 | BLACKHAWK MODIFICATIONS Inc. | EASA.IM.A.S.02359 | REV. 2 | FAA STC SA01946LA | BEECH 200 BEECH 200C BEECH 200CT BEECH 200T BEECH A200 BEECH A200C BEECH A200CT BEECH B200 BEECH B200C BEECH B200CT BEECH B200T BEECH C90 BEECH C90A BEECH E90 BEECH F90 | Installation of Ultra Electronics Engine Instruments.Revision 1: Installation of powerplant indication system. This project is to amend EASA of EASA STC No. EASA.IM.A.S.02359. | | - Installation of in accordance with Master Drawing ListNo. 90-000, Rev. C, dated 24. July 2008 or later approved revision.- Operation must be in accordance with Aircraft Flight Manual Supplement No. AFMS 55023-11 Rev B dated 30. Sept 2008 or (...) | 1. Blackhawk Modifications, Inc. STC SA10341SC or SA103645C is a prerequisite for this modification.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other p (...) | 27.05.2009 |
| 10016381 | BLACKHAWK MODIFICATIONS Inc. | EASA.IM.A.S.02705 | REV. 1 | FAA STC SA10560SC | KING AIR C90GTI | Installation of Wing Mounted Auxiliary Fuel Transfer Tank in BeechC90GTI. This project is to amend EASA STC No. EASA.IM.A.S.02705. | CS-23 | Installation of Wing Mounted Auxiliary Fuel Transfer Tank in accordancewith Master Drawing List No. 8007-II-000, Rev. B, dated 13. December2008 or later approved revision.Type Design Data is provided in Master Drawing 8007-00-000, Rev. C,dated 13. D (...) | None | 16.09.2011 |
| 10016403 | BLACKHAWK MODIFICATIONS Inc. | EASA.IM.A.S.02732 | REV. 1 | FAA STC SA10737SC | BEECH 200, 200C BEECH 200T, 200CT, A200, A200C BEECH A200CT, B200, B200C, BEECH B200CT, B200T | Installation of Pratt and Whitney PT6A-61 engines in Beech 200 Series.This project is to amend EASA STC No. EASA.IM.A.S. 02732. | CS-23 | -Installation of Pratt & Whitney PT6A-61 engines in accordance withMaster Drawing List No. 06010050, Rev. B, dated 27. May 2008 or laterapproved revision. -Operation must be in accordance with Aircraft Flight Manual Supplementcontained in Drawing No. (...) | None | 05.07.2010 |
| 10016484 | BLACKHAWK MODIFICATIONS Inc. | EASA.IM.A.S.02839 | | FAA STC SA10824SC ; RICK RITZ | BEECH 200 BEECH 200C BEECH 200CT BEECH 200T BEECH B200 BEECH B200C BEECH B200CT BEECH B200T | Upgrade Super King Air 200, A200 and B200 Series King Airs to Pratt and Whitney Canada PT6A-52 engines. | | Installation of Pratt & Whitney Pratt and Whitney Pratt & Whitney Canada PT6A-52 in accordance with Blackhawk Modifications Inc. Master Drawing List No. 05210050, Rev. IR, dated 15. Nov. 2007 or later approved revision.Operation must be in accordance (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.The i (...) | 25.08.2008 |
| 10016559 | BLACKHAWK MODIFICATIONS Inc. | EASA.IM.A.S.02944 | | FAA STC SA5786SW | CESSNA 425 | Installation of Pratt and Whitney Canada PT6A - 135A engines on Cessna Conquest 425 airplanes. | | - Installation in accordance with FAA STC SA5786SW and Installation Instructions document by Blackhawk ModificationsInc., datedDecember 15, 1998. - FAA approved AFMS, Blackhawk Modifications Inc. document, revised September 7, 1999, or later FAA approved revision. | None. | 16.02.2009 |
| 10031080 | BLACKHAWK MODIFICATIONS Inc. | EASA.IM.A.S.03186 | | FAA STC SA4705SW | BEECH 65-A90, B90, C90 | Upgrade King Air 65-A90, B90 and C90 airplanes with PT6A-28engines | FAR 23 | Installation of Pratt and Whitney PT6A-27 or PT6A-28 engines in ModelC90 in accordance with drawing 02-001, Revision B,dated June 5, 2008 orlater approved revisions.Or PT6A-27 orPT6A-28 engines in Model 65-A90 or B90 in accordance withdrawing 04-001 (...) | Engines can be either type certified Pratt and Whitney PT6A-27 orPT6A-28 engines, or converted PT6A-27 or PT6A-28 engines from PT6A-21engines in accordance with Atlantic Turbines International Inc STCSE02290NY, or converted PT6A-27 or PT6A-28 engine (...) | 26.07.2010 |
| 10031460 | BLACKHAWK MODIFICATIONS Inc. | EASA.IM.A.S.03209 | | FAA STC SA01248SE | PA-31T PA-31T1 | EASA Validation of FAA STC SA01248SE - Installation of two HartzellHC-E4N-3N/D8990SB propellers | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associatedcertificated noise and/or emissions levels of the original product areunchangedand remain applicable to this certificate/approval. | AFM Supplement AFMS-158P, Rev. Instructions for Continued Airworthiness Doc. ICA-150-5c, Rev. Aor later revisions of the above listed documents approved by EASA inaccordance withEASA ED Decision 2004/04/CF (or subsequent revisions ofthisdecision) | none | 17.08.2010 |
| 10031462 | BLACKHAWK MODIFICATIONS Inc. | EASA.IM.A.S.03208 | | FAA STC SA00863SE | PIPER PA-31T PIPER PA-31T1 PIPER PA-31T2 | EASA Validation of FAA STC SA00863SE- Installation of two Pratt &Whitney PT6A-135 or PT6A-135A engines and two HartzellHC-B3TN-3B/T10178B-8R or HC-E4N-3N/D8990SB propellers | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.FAR23.1, 23.21, 23.51, 23.143, 23.149, 23.711, 23.905, 23.929, 23.933,23.930, 23.1027, (...) | AFM Supplements: AFMS-150T-135, Rev. IR, AFMS-150T-135A, Rev. IR,AFMS-151T-135, Rev. IR, AFMS-151T-135A, Rev. IR, AFMS-152T-135A, Rev. 2,all dated 25 February 2009- Instructionsfor Continued Airworthiness Doc. ICA-150-5b, Rev. Aor later revisions (...) | none | 17.08.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|-----------------------------|---|--|---|---|---|---------------|
| 10031945 | BLACKHAWK MODIFICATIONS Inc. | | | FAA STC SA10816SC | 425 | Installation of Ultra Electronics engine torque indicators in Cessna 425in accordance with FAA STC SA10816SC | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval except for 14 CFR Part 23 paragraphs 23.1, 23.29, 23.301, 23.303,23.305, 23.307, 23.36 (...) | Installation Drawing No. 200707-05, Revision IR, dated July24, 2007Airplane Flight Manual Supplement 200707, Revision IR, dated February26, 2008Instruction for Continued Airworthiness, Doc. No. 200707-30, Rev. IR,dated January 22, 2008or later revis (...) | none | 24.09.2010 |
| 10036326 | BLACKHAWK MODIFICATIONS Inc. | | | FAA STC SA01946LA | PA-31T, PA-31T1, PA31T2 | Installation of Powerplant Indication System. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Installation In Accordance with Blackhawk Modifications Master DrawingList ref 200815-000, Rev A, dated 12-07-09.Operation In Accordance with Blackhawk Modifications Aircraft FlightManual Supplement ref 55023-11, Rev C, dated February 23, 2010.Conti (...) | None. | 01.09.2011 |
| 10036469 | BLACKHAWK MODIFICATIONS Inc. | | | FAA SA02357LA | CESSNA 208B | Replacement of the original P<(>&<)-W PT6A-114(A) engine and propellerwith P<(>&<)-W PT6A-42A and 4-blade Hartzell PropellerHC-E4N-3P/D9900(K) and associated equipment in accordance with FAA STCSA02357LA. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval:FAR23, Amdt. 0: 23.21, 23.171, 23.231, 23.303, 23.307, 23.363, 23.471,23.481, 23.483, (...) | Master Drawing List 200803-000, Revision IR, dated June 8, 2011Flight Manual Supplement AFMS 200803, Rev. IR, dated June 22, 2011Instruction of Continued Airworthiness ICA 200803-30, Rev. A, dated May2011or later revisions of the above listed docume (...) | 1. AeroAcoustics Aircraft Systems, Inc., "Aircraft Payload Extender STOLSystem", STC SA01805SE (EASA STC No. 10029821)is required as part ofthis alteration.2. This modification is not compatible with Cessna Model 208B SerialNumber 208B1190, 208B121 (...) | 12.09.2011 |
| 10014731 | BLOM COMPAGNIA GENERALE RIPRESE AER | EASA.A.S.02913 | | | CESSNA 172P REIMS AVIATION CESSNA F-172 | CGR P001-00 "Structure & Exhaust System Modification for Camera Installation". | | AFM Supplement CGR P001-00 ICA | Applicable to S/N 172-7631 only. | 04.06.2008 |
| 10028772 | BLOM COMPAGNIA GENERALE RIPRESE AER | EASA.A.S.03636 | | | PARTENAVIA P68 PARTENAVIA P68C | | | | | 17.07.2009 |
| 10029675 | BLOM COMPAGNIA GENERALE RIPRESE AER | | | | PA 31-350 | GPS Garmin 430 for IFR | EASA CS 23 Amendment 1The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval | AFMS P/N CGR P003-00-SICA P/N CGR P003-00-lor later revisions of the above listed documents approved by EASA | The installation of EASA approved modification EASA.A.C.07903 isrequired. | 12.04.2010 |
| 10042902 | BLOM COMPAGNIA GENERALE RIPRESE AER | | | | 402C | CGR-P005-00 "Installation of an aerial survey camera kit type UltraCamXp for Aerial Work Survey" | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Report Change CGR-P-005-00 Issue 0 dated 6 August 2010, including FlightManual Supplement "Change CGR-P005-00 Issue 0" Rev.0or later revisions of the above listed documents approved by EASA | Registration Marks EC-DKP | 02.01.2013 |
| 10015918 | BLR AEROSPACE, L.L.C. | EASA.IM.A.S.02051 | REV.2 | FAA STC SA01615SE | BEECH 200, 200C BEECH 300, 300LW BEECH B200, 200C BEECH B200GT, B200CGT | Installation of winglets and associated changes in wing andaleronstructures and related system changes on Raytheon Model 200 200C, B200,B200C, 300 and 300LW airplanes.Revision 1adds models B200GT, B200CGT. | The Certification Basis for the original product remains applicable tothis certificate/ approvalThe requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval | Installation in accordance with BLR Aircraft Flight Manual SupplementsReference AFMS-B200-2, rev. or later approved revisions and AFMS-300-2,rev. - or later approved revisions.-Instructions for Continued Airworthiness in accordance withBLRDdocument (...) | None. | 07.01.2011 |
| 10029451 | BLR AEROSPACE, L.L.C. | | REV. 1 | FAA STC SA02054SE | C90A, C90GT, C90GT1 | Installation of winglets and associated changes in wing andaleronstructures and related system changes on Hawker Beechcraft Model C90A,C90GT and C90GT1. | FAA TC/TCDS No. 3A20. | Installation in accordance with BLR Master Drawing List, DocumentReference BLR KA90W-002, rev. J, dated March 02, 2010,or later approvedrevisions.Operation in accordance with BLRAircraft Flight Manual SupplementReference AFMS-90W-1, rev IR, dated Fe (...) | Eligible Serial Numbers are defined in BLR Master Drawing List, DocumentReference BLR KA90W-002, rev. J, dated March 02, 2010, or later approvedrevisions.Structural life limit of the wing structure as expressed in flight hoursare reduced to, 15,590 hours by the installation of this STC. | 18.05.2010 |
| 10038758 | BLR AEROSPACE, L.L.C. | | | FAA STC SA02131SE | KING AIR B200GT AND B200CGT | Installation of a BLR Aerospace Ultimate Performance Package (UPP). | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval Certification basis defined on FAA STC SA02131SEThis certificate/ approval involves a (...) | Installation in accordance with Master Drawing List (MDL) DocumentBLR-KA250-005, Revision A, dated April 18, 2011. Thisinstallationrequires the adjustment of the Stall Warning System per BLR AerospaceInstallation Instructions document number BLR-KA2 (...) | Approval of this change in type design applies to the above-listedaircraft models only. This approval should not be extended to otheraircraft of this model series unless the installer determines that theratioship between this change and any other (...) | 22.03.2012 |
| 10039059 | BLR AEROSPACE, L.L.C. | | REV. 2 | FAA STC SR01865SE | 1- BELL 212 2- BELL 204B, BELL 205A1 | Installation of Dual Tailboom Strakes and FastFin.Revision 1 adds Model 204B and changes the reference to LBA STC RC 1163to EASA STC 10040783.Revision 2 adds additional Bell 212 and 412 FMSS. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.For components and areas affected by the change i.e. Fast Fin(SR01865SE) andDTBS (STC (...) | RFMS-212FF-1 Rev C dated 28 March 2012 - Flight Manual Supplement;RFMS-212FF-2 Rev IR dated 27 September 2012 - Flight Manual supplement;RFMS-212FF-3 Rev IR dated 27 September 2012 - Flight Manual SupplementModel 212 series with Cargo Hook212-706-10 (...) | Day/Night VFR only, flight into known icing conditions prohibited. | 19.11.2012 |
| 10040783 | BLR AEROSPACE, L.L.C. | | | FAA SR00798SE / LBA RC 1163 | 1- 204B, 205A-1 2- 212, 412, 412EP | Installation of Dual Tailboom Strakes. This STC revises andsupersedesLBA STC RC 1163 to add the Bell Helicopter Textron Inc. Model 204B, 412and 412EP. The AB 212 is removed as its not included in the FAA STC. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | RFMS-BH205-1a revision B dated 8 January 2002 - Flight Manual Supplementfor all models;RFMS-BH205-2a revision IR dated 12 March 2004 - Flight Manual Supplementfor Model 205A-1 Performance Improvement;RFMS-BH205-3a revision IR dated 12 March 2004 - F (...) | Day/Night VFR only, flight into known icing conditions prohibited. | 26.07.2012 |
| 10045079 | BLR AEROSPACE, L.L.C. | | | FAA STC SR009535RC-D | 412 AND 412EP | Installation of Fast Fin System (FAA STC SR009535RC-D). | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.For parts not affected by installation of kit 412-705-040, thecertification basis is a (...) | Rotorcraft Flight Manual Supplement for the BLR FastFin System S/N 36087and Subs (RFMS-412FF-1) revision A dated 21 October 2010;Rotorcraft Flight Manual Supplement for the BLR FastFin System S/N 33001to 33107 (RFMS-412FF-3) revision IR dated 22 Oct (...) | S/Ns 33001 to 33213, 36001 and subsequent onlyFast Fin and Tail Boom Strakes must be installed together.Prior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviously in (...) | 29.05.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|-------------------|----------------|--------------------------------|---------------------------------|---|---|--|--|---------------|
| 10017218 | BOGGI S.R.L. | EASA.S.R.01375 | | | EUROCOPTER FRANCE AS 365N3 | KIT MDS365-P001-00, Additional seat fitting for Eurocopter AS365N3 Helicopter in Hems configuration. | | Boggi S.R.L. Master Drawing List and Master Parts List, Kit/P/N MDS365-P-001-00 Technical Report Part 1B, issue 1, dtd.18-10-2007.Installation in accordance with Boggi S.R.L. Installation Service Bulletin Kit P/N MDS365-P-001-00 Technical Report Par (...) | Limited to AS 365N3 S/N 6609 and 6356, previously modified by the kit Bucher P/N ARA AS365N2-AC50. | 04.12.2007 |
| 10033945 | BOGGI S.R.L. | | | | AS 365 N3 | Kit P/N MDS365-P002-00, External rear view mirror. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | - Boggi s.r.l. Rotorcraft Flight Manual Supplement ,Owner Manual,section 4, Edition 2,Rev. 0 dtd 12/01/2011 or later revisions of theabove listed document approved by EASA- Boggi.s.r.l Master Drawing List Kit MDS365-P-002-00 Technical ReportPart 1B, (...) | -as defined in -Boggi s.r.l. Rotorcraft Flight Manual Supplement,Owner Manual,section 4, Edition 2,Rev. 0 dtd 12/01/2011 | 21.02.2011 |
| 10035798 | BOGGI S.R.L. | | | | AS365N3 | Installation of Draeger TI 500 "Globetrotter" incubator on AS 365N3(ref. Change 365-M003-00). | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Boggi s.r.l. Rotorcraft Flight Manual Supplement 365M00300-FMS-01, issue1 dated 24th June 2011or later revisions of theabove listed document approved by EASA- Boggi s.r.l. Master Document List (MDL) 365M00300-Mdl.-02, issue 2 dated14th July<-> 201 (...) | Limited to:-Incubator model Draeger TI 500 P/N TI 500, S/N 002-Helicopter model AS365N3 S/N 6609 & 6356 | 19.07.2011 |
| 10046582 | BOGGI S.R.L. | | | | P.180 AVANTI | Installation of Satcom Antenna | The Certification Basis (CB) for the original product remains applicaleto this certificate/ approval.The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged an (...) | Master Document List 180P00500-Mdl-00r later revisions of the above listed documents approved by EASA. | Applicable to s/n 1066Prior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noadverse effect upon the airworthiness of the product. | 01.10.2013 |
| 10045814 | BOLDER MANUFACTURING L.L.C. | | | FAA STC ST03233CH | 767-300BCF | Installation of Galley F5 | The Certification Basis (CB) for the original product remains applicaleto this certificate/ approval.The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Bolder Manufacturing Master Data List, MDL-BM0804D01M2, Revision B,18/04/2013or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Techni (...) | None | 19.07.2013 |
| 10015497 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01260 | REV. 1 | TCCA STC C-LSA05-294/D | CHALLENGER CL-600-2B16 | Aircraft Completion 9 Passenger Interior Configuration. (ASN 5595) | | Aero Consulting Services Modification Data Summary E262000 Revision N/C or later approved revision.Aero Consulting Services AFM Supplement E262070 Revision N/C or later approved revision. | Restricted to Aircraft Serial Number 5595 | 04.04.2006 |
| 10015529 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01408 | | FAA STC SA581NE | CL-600-2B16 | Replacement of Total Temperature Sensor | | | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.07.2005 |
| 10015564 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01463 | | C-LSA05-181/D, ISSUE 2 | BOMBARDIER BD-100-1A10 (CHALLE) | Installation of Equipment Ladder (Aft Equipment Bay) in accordance with MDL380801, rev. 1 dated 11 June 2005 and compliance report BAS68380004 | | Maintenance Manual Supplement MMS380801 revision IR dated 11 June 2005 | 1. Limited to: S/N 200132. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 23.09.2005 |
| 10015565 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01464 | | C-LSA05-212/D, ISSUE 1 | BOMBARDIER BD-100-1A10 (CHALLE) | Installation of a Satellite Telephone System Interfaced to an existing Flight Telephone System in accordance with MDL BAS68010010 Revision A dated 30 May 2005 and Compliance Report BAS68380005 dated 10/09/2005 | | Maintenance Manual Supplement BAS68010010-SMM-1 Revision N/Cdated 15 April 2005 | 1. This STC is limited to BD-100-1A10 (Challenger 300) SN 20013 | 05.10.2005 |
| 10015567 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01474 | REV 1 | TCCA STC C-LSA06-015/D ISSUE 2 | BOMBARDIER CL-600-2B16 | Aircraft Completion | | Transport Canada STC C-LSA06-015/D Issue 2 dated September 12, 2006Aero Consulting Services Ltd Modification Data Summary E270000 Revision A or later approved revision.Aero Consulting Services Ltd AFM Supplement E270090 Revision N/C or later approved revision | Restricted to ASN 5605 | 27.09.2006 |
| 10015603 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01539 | | TCCA STC C-LSA06-314/D | BOMBARDIER CL-600-2B16 | Aircraft Completion 11 PassengerInterior Configuration | | Aero Consulting Services Modification Data Summary E265000 Revision N/C or later approved revision.Aero Consulting Services AFM Supplement E265090 Revision N/C or later approved revision. | 1. Restricted to Aircraft Serial Number 5599.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 28.11.2005 |
| 10015674 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01654 | REV 1 | TCCA STC C-LSA06-025/D | BOMBARDIER CL-600-2B16 | Aircraft Completion, 10 Passenger Interior. | | Transport Canada STC C-LSA06-025/D Issue 2 dated August 31 2006 or later approved revision.Aero Consulting Services Modification Data Summary E689000 Revision A or later approved revision.Aero Consulting Services AFM Supplement E689090 Revision N/C or later approved revision. | 1. Restricted to ASN 5613.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 26.10.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|-------------------|----------------|--------------------------------|---------------------------------|---|---|--|--|---------------|
| 10015676 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01657 | REV 1 | TCCA STC C-LSA06-023/D ISSUE 3 | BOMBARDIER CL-600-2B16 | Aircraft Completion, 10 Passenger Interior Configuration | | - Transport Canada STC C-LSA06-023/D Issue 3 dated August 25, 2006.- Aero Consulting Services Modification Data Summary E271000 Revision B or later approved revision.- Aero Consulting Services AFM Supplement E271090 Revision N/C or later approved revision. | 1. Restricted to ASN 5608 | 08.09.2006 |
| 10015677 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01658 | REV 1 | TCCA STC C-LSA06-024/D ISSUE 2 | BOMBARDIER CL-600-2B16 | Aircraft Completion, 10 Passenger Interior Configuration | | - Transport Canada STC C-LSA06-024/D Issue 2 dated August 3, 2006.- Aero Consulting Services Modification Data Summary E688000 Revision N/C or later approved revision.- Aero Consulting Services AFM Supplement E688090 Revision N/C or later approved revision. | 1. Restricted to Aircraft Serial Number 5611.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.09.2006 |
| 10015782 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01828 | | TCCA STC C-LSA06-032/D | BD-700-1A11 | Aircraft Completion 12 Passenger Interior Configuration/Validation of TCCA Limited STC C-LSA06-032/D to S/N 9174MDL: E764000 Rev NC or later TCCA approved revisions | | FMS: G-FM25000137 Rev NC or later TCCA approved revisionsIFCA: GC91740001-IFCA | WLAN system is compatible only to IEEE 802.11(b)Use of IEEE802.11(b) compatible TPEDS for interaction with WLAN is subject to operational rules | 28.04.2006 |
| 10015812 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01881 | | TCCA STC SA06-2 | BD-700-1A10 | Main Entry Door Handrail Modification/Validation of TCCA STCSA06-2MDL: BAS6701036 REV NC or later TCCA approved revisionsTCCA SoC: C-05-1053 SA06-2 dated January 23, 2006 | | IFCA: MMS387201 IR or later TCCA approved revisions | -- | 23.01.2007 |
| 10015816 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01887 | | TCCA STC C-LSA06-178/D | BOMBARDIER CL-600-2B16 (CL-604) | Aircraft Completion | | Aero Consulting Services Modification Sata Summary E921000 Revision N/C or later approved revisionAero Consulting Services AFM supplement E021090 Revision N/C or later approved revision | -- | 27.06.2006 |
| 10015843 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01938 | REV. 2 | TCCA STC SA06-14 ISSUE 2 | BD100-1A10 (CHALLENGER 300) | INSTALLATION OF AN Artex ELT c-406-n WITH A NAVIGATION INTERFACE.Revision of the EASA STC IM.A.S.01938 REV 1 in order to delete the structural provisions approved under STC TCCA SA-05-28 as a prerequisite to TCCA STC SA 06-14 for future a/c deliveries | Certification Basis as per EASA TCDS IM.A.080 Issue 3, 26th October 2010 plus: The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated certificated (...) | - Transport Canada Supplemental Type Certificate SA06-14 Issue 2, dated 9th September 2010. Modification Data Summary E618000 Revision D dated 10th September 2009. Compliance Check List E618065 Revision D dated 25th August 2010. Instructions for Contin (...) | Not applicable | 17.03.2011 |
| 10015885 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02003 | | TCCA STC SA03-113 | CL-600-2B16 (604) | Installation of Complete Custom Aircraft Interior | | Transport Canada STC SA03-113Aero Consulting Services Ltd Configuration Definition List D817000 Revision GAero Consulting Services AFM Supplement D817090 Revision N/C or later approved revision | Limited to ASN 5591 only | 17.08.2006 |
| 10015932 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02070 | | TCCA STC C-LSA06-304/D | CL-600-2B16 | Aircraft Completion | | Transport Canada Limited STC C-LSA06-304/D.Aero Consulting Services Modification data Summary F060000 Revision N/C or later approved revision.Aero Consulting Services AFM Supplement F060090 Revision N/C or later approved revision. | Restricted to ASN 5648 | 16.11.2006 |
| 10015933 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02071 | | TCCA STC C-LSA06-243-D ISSUE 1 | BOMBARDIER CL-600-2B16 (604) | Aircraft Completion | | Transport Canada STC C-LSA06-243-D Issue 1.Aero Consulting Services Modification data Summary E933000 Revision N/C or later approved revision.Aero Consulting Services AFM Supplement E933090 Revision N/C or later approved revision. | 1. Restricted to ASN 5644.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 19.09.2006 |
| 10015967 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02118 | | TCCA STC C-LSA06-317/D | BOMBARDIER BD-700-1A10 | Aircraft Completion, 14 Passenger Interior Configuration/Validation of TCCA Limited STC C-LSA06-317/D to S/N 9197MDL: F017000 Rev NC or later TCCA approved revisions | | SoC: dated Jan 31, 2007FMS: G-FM25000146 Rev NC or later TCCA approved revisionsIFCA: GC91970000-IFCA | Refer to the Limitations sections of the required FM SupplementWLAN system is compatible only to IEEE 802.11(b). The use of WLAN is subject to operational rules. | 31.01.2007 |
| 10015999 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02157 | | TCCA STC SA06-88 | CL-600-2B16 | Activation of Runway Awareness Advisory System (RAAS) Functionality. | | TCCA STC SA06-88, dated 16-11-2006 or later approved revision.Aero Consulting Services Modification Data Summary F036000 Revision N/C or later approved revisions.Aero Consulting Services AFM Supplement F036091 Revision N/C or later approved revisions. | | 06.09.2007 |
| 10016047 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02218 | REV. 1 | TCCA STC SA03-113 | CL-600-2B16 | Complete Custom Interior | | Transport Canada STC SA03-113.Aero Consulting Services Configuration Definition List D817000 Revision W dated November 13, 2006.Aero Consulting Services AFM Supplement D817090 Revision N/C or later approved revision. | Restricted to ASN 5636. | 17.01.2007 |
| 10016062 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02240 | | TCCA LSTC C-LSA07-044/D | BOMBARDIER CL-600-2B16 | Customised Interior Completion | | Transport Canada LSTC C-LSA07-044/Aero Consulting ServicesModification Data Summary E964000 Revision n/c or later approved revision.Aero Consulting Services AFM Supplement E9640920 Revision n/c or later approved revision. | Restricted to ASN 5659. | 11.04.2007 |
| 10016068 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02247 | | TCCA LIMITED STC C-LSA06-281/D | BOMBARDIER BD-700-1A11 | Aircraft Completion, 13 Passenger Interior Configuration/Validation of TCCA Limited STC C-LSA06-281/D to S/N 9186MDL: E915000 Rev A or later approved revisions | | SoC: C-06-0742FMS: G-FM25000141 Rev NC or later approved revisionsIFCA: GC91860000-IFCA IR | Refer to the Limitations Sections of the required FM SupplementWLAN system is compatible only to IEEE 802.11(b)The use of WLAN is subject to operational rules | 23.01.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|-------------------|----------------|-------------------------------|------------------------------|---|---|--|---|---------------|
| 10016075 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02258 | | TCCA LIMITED STC C-LSA07-045D | BOMBARDIER CL-600-2B16 | Customized Interior completion - Aircraft 5664. | | Transport Canada Limited STC No. C-LSA07-045D dated 8 June 2007. Aero Consulting Services Modification Data Summary FO61000 revision N/C or later approved revision. Aero Consulting Services Airplane Flight Manual Supplement FO61000 version N/C or later approved revision. | Restricted to Aircraft Serial No. 5664. | 18.07.2007 |
| 10016081 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02265 | | TCCA STC C-LSA07-111/D | BD-100-1A10 (CHALLENGER 300) | AIRCRAFT COMPLETION, 9 PASSENGER INTERIOR CONFIGURATION/EASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA07-111/D Issue 1 dated 11th May 2007, in accordance with Aero Consulting Services Ltd. Modification Data Summary F111000 Revision N/C. | | Transport Canada Limited Supplemental Type Certificate C-LSA07-111/D Issue 1 dated 11th May 2007. Transport Canada Approved Aero Consulting Services Ltd. Airplane Flight Manual Supplement F111000 Revision N/C. Instructions for Continued Airworthiness HC201375000-IFCA-1 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The u (...) | 15.05.2007 |
| 10016119 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02311 | | TCCA STC SA03-113 | CL-600-2B16 | Complete Custom Aircraft Interior. | | Transport Canada STC SA03-113. Aero Consulting Services Configuration Definition List D817000 Revision AD dated 19 January 2007. Aero Consulting Services AFM Supplement D817090 revision N/C or later approved revision. | Restricted to Serial no. 5655 | 26.03.2007 |
| 10016172 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02399 | | TCCA STC C-LSA07-204/D | BD-100-1A10 (CHALLENGER 300) | AIRCRAFT COMPLETION, 9 PASSENGERS INTERIOR CONFIGURATION /EASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA07-204/D Issue 1 dated 7th September 2007, in accordance with Aero Consulting Services Ltd. Modification Data Summary F384000 Revision N/C | | Transport Canada Limited Supplemental Type Certificate C-LSA07-204/D Issue 1, dated 7th September 2007. Transport Canada Approved Aero Consulting Services Ltd. Airplane Flight Manual Supplement F384090 Revision N/C or later Transport Canada (...) | 1. This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20149.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modi (...) | 24.09.2007 |
| 10016208 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02454 | | TCCA STC SA03-113 | CL-600-2B16 | Complete Custom Aircraft Interior. | | TCCA STC SA03-113. Aero Consulting Services Configuration Definition List D817000 Revision P, dated 4-07-2007. Aero Consulting Services AFM D817090 Revision N/C or later approved revisions.. | Restricted to ASN 5628. | 06.09.2007 |
| 10016237 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02501 | | TCCA STC C-LSA07-366/D | BD-100-1A10 (CHALLENGER 300) | AIRCRAFT COMPLETION, 10 PASSENGERS INTERIOR CONFIGURATION /EASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA07-366/D Issue 1 dated 20th December 2007, in accordance with Aero Consulting Services Ltd. Modification Data Summary F576000 Revision N/C. | | Transport Canada Limited Supplemental Type Certificate C-LSA07-366/D Issue 1, dated 20th December 2007. Transport Canada Approved Aero Consulting Services Ltd. Modification Data Summary ref. F576000 dated 20th December 2007. Revision N/C or late (...) | This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20169. | 10.01.2008 |
| 10016238 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02502 | | TCCA STC C-LSA07-320/D | BD-100-1A10 (CHALLENGER 300) | AIRCRAFT COMPLETION, 10 PASSENGERS INTERIOR CONFIGURATION /EASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA07-320/D Issue 1 dated 20th November 2007, in accordance with Aero Consulting Services Ltd. Modification Data Summary F522000 Revision N/C dated 19.11.07 | | Transport Canada Limited Supplemental Type Certificate C-LSA07-320/D Issue 1, dated 20th November 2007. Transport Canada Approved Aero Consulting Services Ltd. Airplane Flight Manual Supplement F522090 Revision N/C or later Transport Canada (...) | This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20159. | 26.11.2007 |
| 10016362 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02684 | | TCCA STC C-LSA08-014/D | BD-100-1A10 (CHALLENGER 300) | AIRCRAFT COMPLETION, 10 PASSENGERS INTERIOR CONFIGURATION /EASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA08-014/D Issue 1 dated 20th February 2008, in accordance with Aero Consulting Services Ltd. Modification Data Summary F586000 Revision N/C. | | Transport Canada Limited Supplemental Type Certificate C-LSA08-014/D Issue 1, dated 20th February 2008. Transport Canada Approved Aero Consulting Services Ltd. Modification Data Summary ref. F586000 dated 20th February 2008. Revision N/C or later Transport Canada approved revision. , | This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20179. | 25.02.2008 |
| 10016423 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02761 | REV. 38 | TCCA STC SA04-112 | BD100-1A10 (CHALLENGER 300) | Customized Interior Completion. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate: CRI D-07 Issue 2 (Deviation Personal Injury Criteria of Side Facing Seats) CRI D-08 Issue 2 (Equivalent (...) | - Aero Consulting Services Ltd. Configuration Definition List ref. D921000 dated 15 April 2013, Revision RO - Instructions for Continued Airworthiness. For aircraft specific applicability of the Instructions for Continued Airworthiness, refer to the C (...) | EASA approved configurations are identified in document ACSC Configuration Definition List ref. D921000. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed (...) | 22.04.2013 |
| 10016476 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02826 | REV. 2 | TCCA STC SA08-35 ISSUE 3 | BD100-1A10 (CHALLENGER 300) | Observer Seat Installation, Model 409-0001 /Installation of an observer seat between the avionics racks, forward of the main passenger cabin. Revision of EASA STC 10016476, Rev 1 due to revised technical data (Modification Data Summary F797000 Rev. D). | Certification Basis as per EASA TCDS IM.A.080 Issue 2, 8th October 2008. The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise (...) | - Transport Canada Supplemental Type Certificate SA08-35 Issue 3, dated 4th November 2010. - Transport Canada Approved Aero Consulting Services Ltd. Modification Data Summary F797000 Revision D, dated 1st November 2010. - Transport Canada Approved Aero C (...) | Observer Seat for crew use only | 21.01.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|----------------------|----------------|-----------------------------|---|---|--|--|---|---------------|
| 10016477 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02828 | REV. 1 | TCCA STC SA08-58 ISSUE 2 | CL600-2B16 (604 VARIANT) | Installation of AVS-470 Tail Camera System. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Aero Consulting Services Modification Data Summary F782001 Revision A- Instruction for Continued Airworthiness Document F396-IFCA Revision B for later revisions of the above listed documents approved by EASA in accordance with the Canada/EU TIP for Airworthiness and Environmental Certification. | For Configuration -2, -3 and -5, the modification defined by SA97-71 must be embodied as a prerequisite. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modifications will introduce no adverse effect upon the airworthiness of the product. (...) | 11.06.2013 |
| 10016493 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02855 | | TCCA C-LSA08-097/D ISSUE 1 | CL-600-2B16 | Installation of Club Seating | | TCCA STC No. C-LSA08-097/D, Issue 1, dated 20 March 2008. Amtech Aeronautical Limited (AAL) Master Data List MDL418501. Revision IR dated 2008 March 18. AAL Instructions for Continued Airworthiness document ICA418501, Revision IR dated 2008 March 18 or later approved revision. | | 05.05.2008 |
| 10016494 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02856 | | TCCA C-LSA07-418/D ISSUE 1 | BOMBARDIER CL-600-2B16 (604) | Installation of EMS High Speed Data System | | Transport Canada STC No C-LSA07-418/D Issue 1 dated 31 December 2007. Bombardier Master Document List BAS63010116, Revision A dated 14 March 2008 or later approved revision. Bombardier Instructions for Continued Airworthiness (IFCA) document BAS6301011 (...) | 1. Restricted to ASN 54642. Limitation: This approval is limited to 1,500 flight cycles or 12 months from the approval date, whichever comes first until the Airworthiness Limitations subdocument BAS63010116-AWL-1 of Bombardier IFCA BAS63010116-IFCA-1 (...) | 28.07.2008 |
| 10016502 | BOMBARDIER AEROSPACE | EASA.IM.A.S.02865 | | TCCA STC C-LSA08-237/D | BD-100-1A10 (CHALLENGER 300) | CUSTOMIZED INTERIOR COMPLETION # AIRCRAFT 2021210 passenger seats interior configuration. EASA Validation of: Transport Canada Limited Supplemental Type Certificate C-LSA08-237/D Issue 1 dated 24th September 2008, in accordance with Aero Consulting Service (...) | | Transport Canada Limited Supplemental Type Certificate C-LSA08-237/D Issue 1, dated 24th September 2008. Transport Canada Approved Aero Consulting Services Ltd. Modification Data Summary ref. F925000 dated 24th September 2008, Revision N/ (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. (...) | 09.10.2008 |
| 10016631 | BOMBARDIER AEROSPACE | EASA.IM.A.S.03068 | | TCCA STC C-LSA08-275/D ISS1 | BD-100-1A10 (CHALLENGER 300) | Customized Interior Completion - Aircraft 200659 passenger seats interior configuration. EASA Validation of: Transport Canada Limited Supplemental Type Certificate C-LSA08-275/D Issue 1 dated 21st October 2008, in accordance with Aero Consulting Service (...) | | Transport Canada Limited Supplemental Type Certificate C-LSA08-275/D Issue 1, dated 21st October 2008. Transport Canada Approved Aero Consulting Services Ltd. Modification Data Summary ref. G006000 dated 21st October 2008, Revision N/C or later Transport Canada approved revision. Transporto | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. (...) | 06.11.2008 |
| 10016657 | BOMBARDIER AEROSPACE | EASA.IM.A.S.03117 | | TCCA C-LSA09-0002/D ISSUE 1 | BOMBARDIER CL-600-2B16 | Main & Nose Wheel Storage Modifications. | | TCCA STC No. C-LSA09-002/D Issue 1 dated 15 January 2009. Aero Consulting Services Modification Data Summary Number G150000 Revision N/C. Instructions for Continued Airworthiness G150070-ICA Revision N/C. Or any later revisions of the above listed documents approved by EASA in accordance with the Canada/EU TIP for Airworthiness and Environmental Certification. | 1. Restricted to ASN 5426.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the product. (...) | 20.01.2009 |
| 10016671 | BOMBARDIER AEROSPACE | EASA.IM.A.S.03161 | | TCCA STC SA09-16 | BD-100-1A10 (CHALLENGER 300) | Installation of KGS Electronic SS200 Static Inverter. EASA Validation of: Transport Canada Supplemental Type Certificate SA09-16, Issue 2 dated 19th March 2009 in accordance with Aero Consulting Services Ltd. Modification Data Summary G072000 Revision B. | | Transport Canada Supplemental Type Certificate SA09-16, Issue 2 dated 19th March 2009. Aero Consulting Modification Data Summary, ref. No. G072000 Revision B, dated 26th February 2009 or later Transport Canada approved revision. Instructions for Continued Airworthiness, ref. No. G072071-I | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. (...) | 06.04.2009 |
| 10016673 | BOMBARDIER AEROSPACE | EASA.IM.A.S.03176 | | TCCA STC SA04-69 ISSUE 1 | BOMBARDIER BD-100-1A10 (CHALLENGER 300) | Installation of 115 VAC TIA Wavejet Microwave Oven in Forward RH Galley. EASA Validation of: Transport Canada Supplemental Type Certificate SA04-69, Issue 1 dated 4th June 2004 in accordance with Aero Consulting Services Ltd. Modification Data Summary E027000 Revision N/C. | | Transport Canada Supplemental Type Certificate SA04-69, Issue 1 dated 4th June 2004. Aero Consulting Modification Data Summary, ref. No. E027000 Revision N/C, dated 28th May 2004 or later Transport Canada approved revision. Bombardier Aer (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. (...) | 04.05.2009 |
| 10016676 | BOMBARDIER AEROSPACE | EASA.IM.A.S.03189 | | FAA STC ST02683CH | BOEING 747-400 SERIES | Installation of EZMount (TM) mechanical mounting assembly on lower frame of cockpit sliding windows in accordance with Electronic Cable Specialists Master Data List ECS-206728 Rev. E, dated April 8, 2009, or later FAA approved revisions. Validation of FAA STC ST02683CH | | AFM Supplement as listed in Appendix A of ECS Master Data List ECS-206728 Rev. E | 1. AFM Supplement as listed in Appendix A of Electronic Cable Specialists Master Data List ECS-206728, Rev. E, dated April 8, 2009 or later FAA approved revision, is required on board the modified aircraft. The equipment for which these provisions (...) | 18.06.2009 |
| 10027267 | BOMBARDIER AEROSPACE | P. EASA.IM.A.S.03252 | | TCCA STC SA09-43 | BD-100-1A10 (CHALLENGER 300) | VHF COMM 3 Power Source Relocation. EASA Validation of: Transport Canada Supplemental Type Certificate SA09-43, Issue 1 dated 20th May 2009 in accordance with Aero Consulting Services Ltd. Modification Data Summary F912000 Revision N/C. | Certification Basis as per EASA TCDS IM.A.080 Issue 2, 8th October 2008 (JAR 25 Change 15 plus other airworthiness requirements, Special Conditions and Equivalent Safety Findings). | Transport Canada Supplemental Type Certificate SA09-43, Issue 1 dated 20th May 2009. Aero Consulting Modification Data Summary, ref. No. F912000 Revision N/C, dated 12th May 2009 or later Transport Canada approved revision. Instructions for Continued Airworthiness (...) | None | 17.09.2009 |
| 10027876 | BOMBARDIER AEROSPACE | | | TCCA STC SA09-53 ISSUE 1 | CL-600-2B16 | Optional Fuel Quantity Display in Kilograms | EASA IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. SA09-53 Issue 1 dated 29 June 2009. Aero Consulting Services Ltd. Modification Data Summary G336000 Revision N/C or later approved revision or later revisions of the above listed documents approved by EASA in accordance with the Canada/EU TIP for Airworthiness and Environmental Certification. | Not Applicable. | 11.11.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|---------------------|----------------|-----------------------------|---------------------------------------|--|---|--|--|---------------|
| 10028730 | BOMBARDIER AEROSPACE | | | TCCA LSTC C-LSA08-338/D | CL-600-2B16 | Installation, Complete Custom Aircraft Interior - ASN 5506 | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -TCCA LSTC No. C-LSA08-338/D Issue 1 dated 18 December 2008.-Aero Consulting Services Ltd. Modification Data Summary F962000Revision N/C or later approved revision.-Transport Canada approved Aero Consulting Services Ltd. Airplane Flight Manual Supplement (...) | Restricted to ASN 5506. Aircraft must be fitted with an approved means to disconnect the cabin AC and DC utility buses as a prerequisite to this modification. | 05.02.2010 |
| 10029282 | BOMBARDIER AEROSPACE | | | TCCA STC SA10-21 | CL-600-2B16 | Installation, Un-Upholstered Headliner, Rockwell Collins HUD. | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA10-21 Issue 1 dated 04 March 2010.-ACS-NAI Ltd. Modification Data Summary Number G523000 Revision NC or later approved revision. For Instructions for Continued Airworthiness, refer to ACS-NAI Ltd. G523070-ICA Revision NC or la (...) | None. | 17.03.2010 |
| 10029957 | BOMBARDIER AEROSPACE | | | TCCA STC SA10-36 | CL-600-2B16 | Installation, Quick Access Recorder (QAR) | EASA TCDS No. IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Transport Canada STC No. SA10-36 Issue 1 dated 19 April 2010.-ACS-NAI Ltd. Modification Data Summary G551000 Rev N/C, dated 19 April 2010, or later approved revision.-For Instructions for Continued Airworthiness, Refer to Bombardier document TE-31300 (...) | None | 06.05.2010 |
| 10030620 | BOMBARDIER AEROSPACE | | | | BD-100-1A10 (CHALLENGER 300) | Installation of Miniature Quick Access Recorder | Certification Basis as per EASA TCDS IM.A.080 Issue 2, 8th October 2008plus. The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada Supplemental Type Certificate SA10-37 Issue 1, dated 22nd April 2010.-Transport Canada Approved Aero Consulting Services Ltd. Modification data Summary ref. G293000 dated 22nd April 2010, Revision N/C.-Instructions for Continued Airwor (...) | None | 28.06.2010 |
| 10030934 | BOMBARDIER AEROSPACE | | REV. 1 | TCCA STC SA10-50 | BD100-1A10 | Activation of Wireless Local Area Network Access Point on Bombardier BD-100-1A10 (Revision of the EASA STC 10030934, EASA Validation of TCAA approved Supplemental Type Certificate SA10-50 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada Supplemental Type Certificate SA10-50 Issue 1, dated 08th June 2010.- Transport Canada approved ACS-NAI Ltd. Modification Data Summary Number G762000, Rev. A, dated 12th August 2010.- Transport Canada approved ACS-NAI Ltd. Airplane F (...) | None | 23.09.2010 |
| 10030938 | BOMBARDIER AEROSPACE | | REV. 1 | TCCA STC SA10-44 | BD100-1A10 CHALLENGER 300 | Installation of Mechanical Provisions and LRU Installation, Aircell/Air-to-Ground Communication System and Inmarsat Satcom System. Revision of the EASA STC 10030938 (->-) validation of TCAA approved Supplemental Type Certificate SA10-44 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada Supplemental Type Certificate SA10-44 Issue 1, dated 19th May 2010.- Transport Canada approved ACS-NAI Ltd. Modification Data Summary Number G573000, Rev. A, dated 16th August 2010.- Instructions for Continued Airworthiness as per A (...) | None | 23.09.2010 |
| 10033558 | BOMBARDIER AEROSPACE | | REV. 1 | TCCA STC SA10-109 | CL-600-2B16 | Dual Power for Electronic Flight Bag (EFB) - Challenger 604/605 | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA10-109 Issue 1 dated 22 December 2010.-ACS-NAI Ltd. Modification Data Summary Number G918000 Revision A dated 30 May 2011 or later approved revision.-ACS-NAI Ltd. Airplane Flight Manual Supplement G918090 Revision A or late (...) | The complete custom aircraft interior defined by EASA STC No. 10016019 must be installed concurrently or as a pre-requisite. | 11.08.2011 |
| 10033889 | BOMBARDIER AEROSPACE | | | TCCA STC NO. SA11-1 ISSUE 1 | CL-600-2B16 | Activation of Wireless Local Area Network Access Point ICG NXT Mail Server System | EASA.IM.A.023The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. EASA Generic WLAN CRI No. F-01. The requirements for environmental protection (...) | TCCA STC No. SA11-1 Issue 1 dated 04 January 2011.-ACS-NAI Ltd. Modification Data Summary Number G975000 Revision N/C dated 04 January 2011 or later approved revision. ACS-NAI Ltd. Airplane Flight Manual Supplement G975090 Revision N/C or later approved revision (...) | Structural and electrical provisions must be installed in accordance with EASA STC No. 10016019 (TCCA STC No. SA06-90) | 16.02.2011 |
| 10034212 | BOMBARDIER AEROSPACE | | | TCCA STC SA10-93 | CL-600-2B16 | WLAN Activation & Incorporation of Aircell Components | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA10-93 Issue 1 dated 27 October 2010.-ACS-NAI Ltd. Modification Data Summary Number G904000 Revision N/C or later approved revision. ACS-NAI Ltd. Airplane Flight Manual Supplement G904090 Revision N/C or later approved revision (...) | Structural and electrical provisions must be installed in accordance with EASA STC No. 10016019 (TCCA STC No. SA06-90) concurrently or as a pre-requisite. | 16.03.2011 |
| 10036042 | BOMBARDIER AEROSPACE | P-EASA.IM.A.S.03247 | | TCCA STC SA07-110 | BOMBARDIER BD-100-1A10 CHALLENGER 300 | Installation of AVS-470 Tail Camera System. EASA Validation of Transport Canada Supplemental Type Certificate SA07-110. Issue 1 dated 3rd August 2007 in accordance with Aero Consulting Services Ltd. Modification Data Summary F161001. | | - Transport Canada Supplemental Type Certificate SA07-110, Issue 1 dated 3rd August 2007.- Aero Consulting Services Modification Data Summary, ref. No. F161001 Revision B, dated 13th July 2009 or later Transport Canada approved revision.- Instructions for (...) | | 23.07.2009 |
| 10037113 | BOMBARDIER AEROSPACE | | REV. 1 | TCCA STC SA11-41 | BD-100-1A10 (CHALLENGER 300) | Activation of Wireless Local Area Network Access Point, ICG NXT Mail Server System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI F-01 "Installation of Wireless System", Issue 2 dated 29/09/2011. The requirements f (...) | -Modification Data Summary H11000 rev N/C dated 8 June 2011.-Instructions for Continued Airworthiness ref. H110070-ICA rev N/C.- Airplane Flight Manual Supplement ref. H110090 rev N/C or later revisions of the above listed documents approved by EASA inacc (...) | This system is intended to provide cached internet content and e-mail services to Portable Electronic Devices. Any other use of this equipment will require a re-examination of the certificate basis. This STC is a design certification approval for the act (...) | 07.11.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|------------------------------|------------------------|---|--|---|--|---------------|
| 10037705 | BOMBARDIER AEROSPACE | | | TCCA STC NO. SA11-89 ISSUE 1 | CL-600-2B16 | Modification to Potable Water Tank Air Supply | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged (...) | TCCA STC NO. SA11-89 Issue 1 dated 07 November 2011. ACS-NAI Ltd Modification Data Summary Number H311000 Revision N/C or later approved revision. Instructions for Continued Airworthiness. Bombardier document BAS63058003 Revision NC or later approved rev (...) | Previous installation of a specific potable water system is required. Drawings listed in Modification Data Summary H311000 identify the prerequisite installations. | 16.12.2011 |
| 10038288 | BOMBARDIER AEROSPACE | | | TCCA STC SA11-88 | CL-600-2B16 | Electronic Flight Bag Class II (PED) (Provisions) | IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged (...) | -TCCA STC NO. SA11-88 Issue 1, dated 25 October 2011. ACS-NAI Ltd. Modification Data Summary Number H220000 Revision N/C, dated 25 October 2011 or later approved revision. ACS-NAI Ltd. Airplane Flight Manual Supplement H220090 Revision N/C or later approved revision (...) | None | 03.02.2012 |
| 10038597 | BOMBARDIER AEROSPACE | | | TCCA STC NO. SA12-12 ISSUE 1 | CL-600-2B16 | Galley Modification for Microwave Overheat | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged (...) | TCCA STC NO. SA12-12 Issue 1 dated 23 January 2012. ACS-NAI Ltd Modification Data Summary Number H372000 Revision N/C dated 23 January 2012 or later approved revision. Instructions for Continued Airworthiness, ref. to Bombardier document BAS63058004 Rev (...) | As a prerequisite aircraft must be fitted with EASA STC No. 10016019. Installation of Complete Custom Aircraft Interior. | 29.02.2012 |
| 10042975 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01220 | | | BD700-1A10 | Cabin Interior Completion. | | Doc. Limited Supplemental Type Certificate by TCCA No. C-LSA05-103/DFMS: G-FM25000126 N/CALI: G-TM05100065 N/C, or later approved revisions | 1. This STC is limited to Serial Number 9146.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 14.06.2005 |
| 10042979 | BOMBARDIER AEROSPACE | EASA.IM.A.S.01505 | | TCCA STC C-LSA06-002/D | BD-100-1A10 | Cabin mod for passenger seat upgrade in accordance with Bombardier MDLBA68010009 | | None | 1. SN 20013, validation of TCCA STC C-LSA06-002/D2. Instructions for continuing airworthiness and supplemental maintenance information are provided in Amtech Aeronautical Limited documents MMS390201.3. Bombardier Aerospace service bulletin No. 100-53- (...) | 04.05.2006 |
| 10043158 | BOMBARDIER AEROSPACE | EASA.IM.A.C.01241 | | | | Mod Sum (88) & 100T01012/Rockwell Collins V 6.0 Avionics Part Number Update. | | AFM RS-4 previously approved under approval 2005-4140 on May 10, 2005 is associated to this change | None | 03.05.2006 |
| 10045060 | BOMBARDIER AEROSPACE | | | TCCA SA04-102 ISSUE 1 | BD-100-1A10 | Activation, Rockwell Collins 5000/6000 SATCOM | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Modification Data Summary Number E026000 Revision A or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/ Canada Bilateral Agreement. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 28.05.2013 |
| 10045439 | BOMBARDIER AEROSPACE | | | C-LSA07-156/D | BD-700-1A10 | Aircell ST3100 Satcom Installation | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. C-LSA07-156/D, Issue 2 dated 19 December 2008. Bombardier Aerospace Master Document List BAS67017018, Revision B dated 24 October 2008. Bombardier Aerospace Instructions for Continued Airworthiness Document BAS67057020, Revision (...) | Restricted to ASN 9003/Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the air (...) | 25.06.2013 |
| 10045899 | BOMBARDIER AEROSPACE | | | TCCA P-LSA05-163/D | BD-100-1A10 | Installation of a Digital Voice Check-List Management System in accordance with Master Document List BAS68010006 rev A dated November 3, 2005 | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Bombardier Business Aviation Services BBAS Master Document List BAS68010006 rev A dated November 3, 2005- Amtech Aeronautical Limited (AAL) Flight Manual Supplement FMS388401 rev IR dated 28 October 2005- Instructions for Continued Airworthiness and (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 26.07.2013 |
| 10046730 | BOMBARDIER AEROSPACE | | | TCCA STC C-LSA05-376/D | BD100-1A10 | Cabin Modification for Passenger Seat Upgrade in accordance with Master Document List BAS69010005 rev A dated November 3, 2005. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Bombardier Business Aviation Services BBAS Master Document List BAS69010005 rev A dated November 3, 2005- Amtech Aeronautical Limited (AAL) document MMS387101, Revision IR dated 2005 November 1 or later revisions of the above listed documents approve (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 16.10.2013 |
| 10015690 | BOMBARDIER AEROSPACE/LEARJET INC | EASA.IM.A.S.01677 | | FAA STC ST03023AT | BD-700-1A10 | Installation of a Lower Fuselage Cantilever Drain Mast Validation of FAA STC ST03023AT as Basic STC MDL: BAS67017009 | | FAA Statement of Compliance (SoC) dated Mar 17, 2006 | | 06.04.2006 |
| 10015592 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.01504 | REV. 1 | TCCA STC C-LSA05-272/D | BOMBARDIER BD-700-1A10 | 1. Installation, Aircraft Interior Completion for BD700-1A10 Global Express acc MDS E553000 Rev N/C.2. Revision 1: Modification to Interior Completion. | | 1. AFM Supplement G-FM25000132 Rev N/C or later approved revision.2. ICA GC91590001-IFCA.3. AMM Supplement/Airworthiness Limitations G-TM05100068.4. Revision 1:4.1. Aero Consulting Services Ltd Modification Data Summary No E553000 Revision A dated 16 October 2008 or later approved rev | 1. Ref G-FM25000132 Rev N/C Limited to S/N 9159.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations (...) | 26.11.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|--------------------------------|--|---|---------------------|--|---|---------------|
| 10015955 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02098 | | C-LSA06-256/D | BOMBARDIER BD-700-1A11 | Aircraft Completion, 13 Passenger Interior Configuration/Validation of TCCA Limited STC C-LSA06-256/D to S/N 9188MDL: E786000 Rev NC or later TCCA approved revisions | | SoC: C-06-0642, dated Sep 28, 2006FMS: G-FM25000140 Rev NC or later TCCA approved revisionsIFCA: GC91880000-IFCA | Refer to the Limitations sections of the required FM Supplement WLAN system is compatible only to IEEE 802.11(b). | 04.10.2006 |
| 10015957 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02100 | | TCCA STC C-LSA06-291/D | BOMBARDIER BD-700-1A10 | Aircraft Completion, 12 Passenger Interior Configuration/Validation of TCCA Limited STC C-LSA06-291/D to S/N 9189MDL: F015000 Rev NC or later approved revisions | | SoC: C-06-0611, dated Nov 24, 2006FMS: G-FM25000143 Rev NC or later approved revisionsIFCA: GC91890000-IFCA IR | Refer to the Limitations sections of the required FM Supplement WLAN system is compatible only to IEEE 802.11(b).The use of WLAN is subject to operational rules. | 30.11.2006 |
| 10015958 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02101 | | TCCA STC C-LSA06-336/D | BOMBARDIER BD-700-1A11 | Aircraft Completion, 13 Passenger Interior Configuration/Validation of TCCA Limited STC C-LSA06-336/D to S/N 9204MDL: F062000 Rev NC or later approved revisions | | SoC: C-06-0896, dated Nov 24, 2006FMS: G-FM25000145 Rev NC or later approved revisionsIFCA: GC92040000-IFCA IR | Refer to Limitation sections of the required FM Supplement | 05.12.2006 |
| 10015959 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02102 | | C-LSA06-321/D | BOMBARDIER BD-700-1A10 | Aircraft Completion, 14 Passenger Interior Completion/Validation of TCCA Limited STC C-LSA06-321/D to S/N 9200MDL: F099000 Rev NC or later TCCA approved revisions | | SoC: C-06-0854, dated Dec 18, 2006FMS: G-FM25000147 Rev NC or later TCCA approved revisionsIFCA: GC92000000-IFCA IR | Refer to the Limitations sections of the required FM Supplement WLAN system is compatible only to IEEE 802.11(b).The use of WLAN is subject to operational rules. | 20.12.2006 |
| 10015960 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02103 | | TCCA STC C-LSA07-015/D | BOMBARDIER BD-700-1A10 | Customized Interior Completion - SN 9196.Validation of TCCA STC C-LSA07-015/D. | | AFM Supplement G-FM25000148.TCCA SoC letters: C-06-0687 dated November 09, 2007.Instructions for Continuous Airworthiness : GC91960000-IFCA. | Refer to the Limitations sections of the required AFM Supplement.WLAN system is compatible only to IEEE 802.11(b).The use of WLAN is subject to operational rules. | 29.11.2007 |
| 10015961 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02106 | | TCCA STC C-LSA07-013/D | BOMBARDIER BD-700-1A10 | Aircraft Completion, 13 Passenger Interior Configuration/Validation of TCCA Limited STC C-LSA07-013/D to S/N 9202MDL: F018000 Rev NC or later TCCA approved revisions | | SoC: C-06-0613 E786 dated Mar 01, 2007FMS: G-FM25000151 Rev NC or later TCCA approved revisionsIFCA: GC92020000-IFCA | Refer to the Limitations sections of the required FM Supplement WLAN system is compatible only to IEEE 802.11(b).The use of WLAN is subject to operational rules. | 21.03.2007 |
| 10016011 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02176 | | | BOMBARDIER BD-700-1A11 | Aircraft Completion, 13 Passenger Interior Configuration/Validation of TCCA Limited STC C-LSA06-335/D to S/N 9198MDL: F069000 Rev NC or later approved revisions | | SoC: C-06-0891, dated Nov 24, 2006FMS: G-FM25000144 Rev NC or later approved revisionsIFCA: GC91980000-IFCA IR | Refer to Limitation sections of the required FM Supplement WLAN system is compatible only to IEEE 802.11(b).The use of WLAN is subject to operational rules. | 05.12.2006 |
| 10016055 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02227 | | TCCA STC C-LSA07-086/D REV NC | BOMBARDIER BD-700-1A10 BOMBARDIER BD-700-1A11 | Customized Interior Completion - SN 9208/Validation of TCCA STC C-LSA07-086/D Rev NC | | AFM Supplement G-FM25000154TCCA SoC letter C-06-0614, dated June 12, 2007/ICA: GC92080000-IFCA | 1. Refer to the Limitations sections of the required FM Supplement2. WLAN system is compatible only to IEEE 802.11(b).3. The use of WLAN is subject to operational rules. | 21.06.2007 |
| 10016136 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02332 | | TCCA STC C-LSA07-087/D ISSUE 1 | BOMBARDIER BD-700-1A10 | Customized Interior Completion - SN 9213/Validation of TCCA STC C-LSA07-087/D issue 1 | | AFM Supplement G-FM25000155.TCCA SoC letter C-06-0668, dated June 19, 2007/ICA: GC92130000-IFCA | Refer to the Limitations sections of the required FM Supplement.WLAN system is compatible to IEEE 802.11(b).The use of WLAN is subject to operational rules.Prior to installation of this modification the installer must determine that the interrelatio (...) | 09.07.2007 |
| 10016214 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02462 | | TCCA STC C-LSA02-251/D | BOMBARDIER BD-700-1A10 | Customized Interior Completion - SN 9084/Validation of TCCA STC C-LSA02-251/D issue 2 | | AFM Supplements G-FM2500029A / BTCCA : SoC letters C-03-0390 dated Sep 26 and Oct 04, 2007Time limits and Maintenance Checks : G-TM05200018 | Refer to the Limitations sections of the required AFM Supplement | 05.10.2007 |
| 10016242 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02515 | | TCCA STC SA05-60 | BOMBARDIER BD-700-1A11 | Customized Interior Completion - SN 9182.Validation of TCCA STC SA05-60.via SoC limited to SN 9182.Configuration acc: CDL E25000 Rev N | | AFM Supplements: see TCCA STC SA05-60TCCA SoC letter C-06-0571 dated Oct 12, 2007Time limits and Maintenance Checks : see TCCA STC SA05-60 | Refer to the Limitations sections of the required AFM Supplement | 24.10.2007 |
| 10016259 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02538 | REV. 1 | TCCA SA04-113 ISSUE 1 (CDL REV | BD-700-1A10 | Cabin Interior Door Modifications - ASN 9167 | EASA.IM.A.009 | EASA STC No. EASA.IM.A.S.02538 dated 18 October 2007.TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.TCCA SoC ref. C-09-0419 dated 18 June 2009.Aero Consulting Services Ltd Configuration Definition List No. 0954000Revision CX :. dated 01 May 200 (...) | Restricted to ASN 9167. | 20.12.2007 |
| 10016301 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02598 | | TCCA STC C-LSA08-019/D | BOMBARDIER BD-700-1A11 | Customized interior Completion - ACSN 9257 | | 1. Transport Canada STC No. C-LSA08-019/D Issue 1 dated February 22 2008. Transport Canada Statement of Compliance C-07-0007 dated 12 March 2008/2. Aero Consulting Services Modification Data Summary F528000 Revision N/C.3. Airplane Flight M (...) | 1. Restricted to ASN 9257/2. WLAN System in only compatible to IEEE802.11(b).3. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed (...) | 20.03.2008 |
| 10016302 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02599 | | TCCA LIMITED STC C-LSA07-284/D | BOMBARDIER BD-700-1A10 | Customized Interior Completion - ACSN 9235 | | TCCA STC C-LSA07-284/D, Issue 1, dated 30-11-2007.TCCA SOC letter C-07-0765, dated 7-12-2007.Aero Consulting Services Modification Data Summary F460000 Revision N/C dated 30-11-2007.Bombardier Airplane Flight Manual Supplement G-FM25000165, Revision (...) | Refer to the limitations section of the required AFM Supplement and to the referenced FCOMS.WLAN system is compatible only to IEEE 802.11 (B). | 21.12.2007 |
| 10016303 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02600 | | TCCA STC C-LSA07-286/D FOR CAB | BOMBARDIER BD-700-1A10 | Customised Interior Completion -ACSN 9238. | | TCCA SOC letter C-07-0767 dated 17 January 2008.Aero Consulting Services Modification Data Summary F462000 Revisoin N/Cdated 10 December 2007.Bombardier Airplane Flight Manual Supplement G-FM25000164 Revision N/C.Instructions for Continued Airworthiness GC92380000-IFCA. | Restricted to ASN 9238 and Refer to the limitations sections of the required AFM supplement and to the referenced FCOMS.WLAN System is compatible only to IEEE 802.11(b). | 21.01.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|--------------------------------|---------------------------------|--|---------------------|--|---|---------------|
| 10016305 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02602 | | TCCA STC (SA05-60) FOR CABIN | BOMBARDIER BD-700-1A11 | Customized Interior Completion - ACSN 9241 | | TCCA SOC letter C-08-365 dated 30 April 2008.Aero Consulting Services Configuration Definition List E025000 Revision AKdated 29 November 2007.Bombardier Airplane Flight Manual Supplement G-FM25004051 Revision N/C or later approved revision.Instruction for continued Airworthiness GC92410000-IFCA. | Restricted to ASN 9241 and 1. Refer to the Limitations sections of the required AFM Supplement. | 07.05.2008 |
| 10016306 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02603 | | TCCA LIMITED STC C-LSA07-278/D | BOMBARDIER BD-700-1A11 | Customized Interior Completion - ACSN 9243. | | TCCA STC C-LSA07-278/D, Issue 1, dated 05-12-2007.TCCA SOC letter C-07-0704 dated 7-12-2007.Aero Consulting Services Modification Data Summary F4570000 Revision N/C dated 06-12-2007.Bombardier Airplane Flight Manual Supplement G-FM2500162, Revision (...) | Refer to the limitations section of the required AFM Supplement and to the referenced FCOMS.WLAN system is compatible only to IEEE 802.11 (B). | 21.12.2007 |
| 10016311 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02614 | | TCCA STC SA 03-113 | BOMBARDIER CL-600-2B16 (604) | Installation, complete Custom Interior | | Transports Canada STC SA 03-113 Issue 1 dated December 19 2003 Aero Consulting Services Ltd. Configuration Definition List D817000 Revision A1.Aero Consulting Services Ltd AFM SupplementD817090 Revision A or later approved revision. | Restricted to ASN 5623. | 02.11.2007 |
| 10016330 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02639 | | TCCA 0-LSA01-007 ISSUE 6 | BOMBARDIER BD-700-1A10 | Interior and Avionics Installation. | | TCCA SoC 5010-0-07-0746 dated 13 May 2008Deca Aviation Engineering Ltd MDL No. MDL9004, Revision E, dated 10 April 20021. The following DECA Aviation Engineering Ltd Flight Manual Supplements! are required:A. interior and Avionics I n sial lai (...) | 1. Limited to AS N90042. Compliance with Bombardier Aerospace Airworthiness Limitations GC-90045000-AWL, dated July 15, 2002, is mandatory.3. Occupancy# Maximum numberof passengers Twelve (12)# Maximum number of occupants including (...) | 22.05.2008 |
| 10016355 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02676 | | TCCA O-LSA07-261/D ISSUE 1 07/ | BOMBARDIER BD-700-1A10 | Cabine Floor Structure Modification & Decompression Panel Relocation. | | TCCA Statement of compliance reference 5010-O-07-0790 dated24 April 2008.DECA Aviation Modification Summary MS07159 Revision 1 dated 18 September 2007 or later approved revision.Instruction for Continued Airworthiness ICA07159 Revision 1 dated 22 October 2007 or later approved revision. | | 05.05.2008 |
| 10016377 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02701 | | TCCA STC SA04-113 | BOMBARDIER BD-700-1A10 | Customized interior Completion - ACSN 9254. | | TCCA SOC Letter C-08-0342 dated 06 May 2008.Aero Consulting Services Configuration Deviation List D954000 Revision BK dated 11 April 2008.Bombardier Airplane Flight Manual Supplement G-FM25004001 Revision N/C.Instruction for Continued Airworthiness GC92540000-IFCA. | Restricted to ASN 9254 and 1. Refer to AFM Supplement and to the reference FCOMS.WLAN system is compatible only to IEEE 802.11(b). | 08.05.2008 |
| 10016382 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02706 | | TCCA STC SA01-156 | BOMBARDIER BD-700-1A10 | Multi-Function Ladder Installation. | | Transport Canada Statement of Compliance ref. NAI 5012-01-0610Transport Canada STC No. SA01-156 Issue 1 dated 10 December 2001Elsen Technologies Inc. Master Data List No. MDL-02GX-117 Revision A dated 20 December 2001 or later approved revision | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 16.07.2008 |
| 10016416 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02749 | | TCCA STC C-LSA08-089/D | BOMBARDIER BD-100-1A10 (CHALLE) | AIRCRAFT COMPLETION, 10 PASSENGERS INTERIOR CONFIGURATION /EASA Validation ofTransport Canada Limited Supplemental Type Certificate C-LSA08-089/D Issue 1 dated 21st April 2008, in accordance with Aero ConsultingServices Ltd. Modification Data Summary F791000 Revision N/C. | | Transport Canada Limited Supplemental Type CertificateC-LSA08-089/D Issue 1, dated 21st April 2008. TransportCanada Approved Aero Consulting Services Ltd. Modification Data Summary ref. F791000 dated 21st April 2008, RevisionN/Cor later Transport Canada approved revision. Transport C | This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20189. | 23.04.2008 |
| 10016436 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02777 | | TCCA C-LSA08-062/D ISSUE 1 | BOMBARDIER BD-700-1A10 | Customized Interior Completion - ACSN 9250. | | TCCA STC No. C-LSA08-062/D Issue 1 dated 26 June 2008.TCCA SOC letter C-08-0147 dated 03 July 2008.Aero Consulting Services Modification Data Summary F632000 Revision N/C dated 26June 2008.Bombardier Airplane Flight Manual Supplement G-FM25000169 Re (...) | Restricted to ASN 9250 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2. WLAN system is compatible only to IEEE 802.11(b).3. Prior to installation of this modification the installer must determine (...) | 17.07.2008 |
| 10016438 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02779 | | C-LSA08-065/D ISSUE 1 | BOMBARDIER BD-700-1A11 | Customized Interior Completion - ACSN 9269 | | Transport Canada STC No. C-LSA08-065/D issue 1 dated 1 May 2008.Transport Canada Statement of Compliance C-08-0157 dated 9 May 2008.Aero Consulting Services Modification Data Summary F731000 Revision N/C.Airplane Flight Manual Supplement G-FM2500017 (...) | Restricted to ASN 9269.WLAN System is only compatible to 1EEE802.11 (b). | 22.05.2008 |
| 10016448 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02794 | | TCCA STC SA04-113 | BOMBARDIER BD-700-1A10 | Customized Interior Completion - ACSN 9262 | | TCCA STC No. SA04-113 Issue 1 dated 28 October2004.TCCA SOCletter C-08-0419 dated 28 May 2008.Aero Consulting ServicesConfiguration Deviation List D954000 Revision BN dated 09 May 2008.Bombardier Airplane Flight Manual Supplement G-FM25004001 Revisi (...) | Restricted to ASN 9262 and 1.Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b).3.Prior to installation of this modification the installer must determine that (...) | 09.06.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|--------------------------|--|--|--|---|--|---------------|
| 10016449 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02795 | | | BOMBARDIER BD-700-1A11 | Customized Interior Completion - ACSN 9273. | | TCCA STC No. SA05-60 Issue 1 dated 14 June 2005.TCCA SOC letter C-08-0626 dated 02 September 2008.Aero Consulting Services Modification Data Summary E025000 Revision AW dated 29 August 2008.Bombardier Airplane Flight Manual Supplement G-FM25004501 R (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 10.09.2008 |
| 10016457 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02803 | | TCCA C-LSA08-081/D | BOMBARDIER BD-700-1A10 | Interior Completion, BD-700-1A10 serial number 9252 | | 1. TCCA STC No.C-LSA08-081/D Issue 1 dated 22 October 2008.2. TCCA SoC reference C-08-0201 dated 30 October 2008.3. Aero Consulting Services Modification Data Summary F745000 Revision N/C or later approved revision.4. Bombardier Flight M (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 07.11.2008 |
| 10016458 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02804 | REV. 3 | TCCA STC NO. SA04-113 | BD-700-1A10 | Customized Interior Completion - ACSN 9272. | EASA TCDS No. IMA.009 - The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this approval. | 1. TCCA SoC letter C-09-0997 dated 24 November 2009.2. AeroConsulting Services Configuration Definition List D954000Revision CA dated 25 November 2008.3. Bombardier Airplane Flight Manual Supplement G-FM25004001 Revision B.4. Instructionsfor Continu (...) | 1. Restricted to ACSN 9272.2. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS.3. WLAN systems are compatible only to IEEE 802.11(b). | 18.12.2009 |
| 10016459 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02805 | | TCCA STC C-LSA08-082/D | BOMBARDIER BD-700-1A10 | Interior Completion, BD-700-1A10 serial number 9256. | | 1. TCCA STC No. C-LSA08-082/D Issue 1 dated 10 October 2008.2. TCCA SOC Reference C-08-0202 dated 23 October 2008.3. Aero Consulting Services Modification Data Summary F746000 Revision N/C'4. Bombardier Flight Manual Supplement G-FM2500 (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 28.10.2008 |
| 10016452 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02920 | | TCCA STC SA04-113 | BOMBARDIER BD-700-1A10 | Interior Completion - ACSN 9234 | | TCCA STC No. SA04-113 Issue 1 dated 29 October 2004.TCCA SOC letter C-07-0590 dated 03 July 2008.Aero Consulting Services Modification Data Summary D954000 Revision AY dated 15 November 2007.Bombardier Airplane Flight Manual Supplement G-FM25004001 (...) | Restricted to ASN 9234 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2. WLAN system is compatible only to IEEE 802.11(b).3. Prior to installation of this modification the installer must determine (...) | 28.07.2008 |
| 10016543 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02921 | | TCCA STC SA06-43 ISSUE 2 | BOMBARDIER BD-700-1A10 BOMBARDIER BD-700-1A11 | Installation of XM Satellite Radio Antenna. | | Transport Canada STC No. S06-43 Issue 2 dated 10 November 2006. Transport Canada Statement of Compliance ref. C-06-0865 dated 21 July 2008.Aero Consulting Services Modification Data Summary no. E321000 Revision C dated 20 October 2006 or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 22.08.2008 |
| 10016565 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02956 | | TCCA STC C-LSA03-458/D | BOMBARDIER BD-700-1A10 | Installation of Thales Integrated Standby Instrument. | | - TCCA SoC ref. C-08-0753 dated 22 December 2008.- TCCA STC No. C-LSA03-458/D Issue 1 dated 30 September 2003.- Aero Consulting Services Modification Data Summary Number D758000 Revision N/C or later approved revision.- Aero Consulting Services Flight Manual Supplement G-FM3400004 Revis | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 23.12.2008 |
| 10016566 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02957 | | TCCA STC C-LSA03-372/D | BOMBARDIER BD-700-1A10 | Aircraft Completion, 13 Passenger Interior Configuration | | Transport Canada SoC ref. C-08-0754 dated 13 November 2008.Transport Canada STC No. C-LSA03-372/D Issue 2 dated 13 November 2008.Aero Consulting Services Modification Data SummaryD583000 Revision A dated 13 November 2008 or later approvedrevision.Re (...) | 1. Restricted to ASN 9120.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airwor (...) | 15.12.2008 |
| 10016567 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02958 | | TCCA STC C-LSA01-417/D | BOMBARDIER BD-700-1A10 | Approval of Side Facing Divan Limitations. | | Transport Canada STC No. C-LSA01-417/D Issue 2 dated 16 January 2004.TCCA SOC reference C-04-0040 dated 16 September 2008.Aero Consulting Services Modification Data Summary No. D14000 Revision A or later approved revision.Aero Consulting Services FI (...) | Restricted to ASN 9016. Required Equipment:Previous installation of corresponding EASA STC No. EASA.IM.A.S.01252 (FAA STC SA8164NM-D).Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 19.09.2008 |
| 10016572 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02970 | | TCCA STC C-LSA03-334/D | BOMBARDIER BD-700-1A10 | Installation of interior furnishings and cabin systems, applicable to BD-700-1A10 serial number 9112 only. | | TCCA STC No. C-LSA03-334/D Issue 2 dated 06 October 2003.TCCA SOC Reference C-08-0755 dated 16 September 2008.Aero Consulting Services Modification Data Summary D42500 Revision B or later approved revision.The following Flight Manual Supplements are (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 02.10.2008 |
| 10016573 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.02971 | | TCCA STC C-LSA05-100/D | BOMBARDIER BD-700-1A10 | Relocation of Aft Toilet Compressor. | | Transport Canada STC No. C-LSA05-100/D Issue 1 dated 17 March 2005.TCCA SOC reference C-08-0755 dated 16 September 2008.Aero Consulting Services Modification Data Summary No. E395000 Revision N/C dated 17 March 2005 or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 19.09.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|---------------------|----------------|---------------------------|--|--|---|--|---|---------------|
| 10016591 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.03001 | | TCCA STC SA04-113 | BOMBARDIER BD-700-1A10 | Customized Interior Completion - ACSN 9270. | | TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.TCCA SOC reference C-08-0433 dated 17 November 2008.Aero Consulting Services Configuration Deviation List D954000 Revision BX dated 14 November 2008.Instructions for Continued Airworthiness GC92700000-IFCA. | Restricted to ASN 9270 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b).3.Prior to installation of this modification the installer must determine th (...) | 21.11.2008 |
| 10016593 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.03004 | REV. 1 | TCCA STC SA07-019 | BOMBARDIER BD-700-1A10 BOMBARDIER BD-700-1A11 | Waterproofing and Spill Management, Interior Modifications | | TCCA STC No. SA07-19 Issue 3 dated 25 July 2008.TCCA SoC reference NA15012-07-0177 dated 27 February 2009.Kmix Aviation Inc. Master Document List - Water Proofing and Spill Management, Interior Modifications - Bombardier BD-700 Aircraft, Document N (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 16.03.2009 |
| 10016635 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.03074 | | TCCA STC SA04-113 ISSUE 1 | BOMBARDIER BD-700-1A10 | Installation, Complete Custom Aircraft Interior / Installation of cabin systems and custom furnishings - ACSN 9291 | | - TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.- TCCA SOC reference C-09-0047 dated 16 January 2009.- Aero Consulting Services Configuration Deviation List D954000Revision CE dated 12 January 2009.- Instructions for Continued Airworthiness GC92910000-IFCA. | 1. Restricted to ASN 9291.2. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.3.WLAN system is compatible only to IEEE 802.11(b).4. Prior to installation of this modification the installer must (...) | 20.01.2009 |
| 10016652 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.03100 | | TCCA STC SA05-60 | BOMBARDIER BD-700-1A11 | Customized Interior Completion - ACSN 9293 | | TCCA STC No. SA05-60 Issue 1 dated 14 June 2005.TCCA SOC letter C-08-0874 dated 18 December 2008.Aero Consulting Services Modification Data Summary E025000 Revision BE dated 12 December 2008.Bombardier Airplane Flight Manual Supplement G-FM25004501 (...) | Restricted to ASN 9293 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2.2. WLAN system is compatible only to IEEE 802.11 (b).3.Prior to installation of this modification the installer must determine (...) | 14.01.2009 |
| 10016653 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.03102 | | | BOMBARDIER BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9286 | | Transport Canada Limited STC No. C-LSA09-003D dated 07 January 2009.Transport Canada SoC reference C-08-0994 dated 21 January 2009.Aero Consulting Services Modification Data Summary F990000 Revision N/C or later approved revision.Bombardier Flight (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 26.01.2009 |
| 10016656 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.03105 | | SA05-60 | BOMBARDIER BD-700-1A11 | Installation, Complete Custom Aircraft Interior | | Transport Canada STC No. SA05-60 Issue 1 dated 14 June 2005.Transport Canada SoC ref. C-09-0200 dated 31 March 2009.Aero Consulting Services Ltd. Configuration Definition List E025000 Revision BG or later approved revision.Aero Consulting Services L (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 02.04.2009 |
| 10016667 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.03140 | | TCCA STC SA06-34 | BOMBARDIER BD-700-1A10 BOMBARDIER BD-700-1A11 | Installation of CMS-400 Digital Voice Checklist System. | | - Transport Canada STC No. SA06-34 Issue 1 dated 23 May 2006.- Transport Canada SoC ref. C-09-0335 dated 15 April 2009.- Aero Consulting Services Ltd. Modification Data Summary no. E938000 Revision A or later approved revision.- Bombardier Aerospace FMS No. G-FM31404001 Revision N/C or l (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 27.04.2009 |
| 10016668 | BOMBARDIER COMPLETION CENTRE INC | EASA.IM.A.S.03141 | | TCCA STC SA04-113 ISSUE 1 | BOMBARDIER BD-700-1A10 | Interior Completion, BD-700-1A10 - ASN 9281 | | Transport Canada SoC reference C-09-0174 dated 09 March 2009.TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.Aero Consulting Services Configuration Definition List D954000 Revision CB dated 11 December 2008.Instructions for Continued Airworthiness GC92810000-IFCA. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 10.03.2009 |
| 10027058 | BOMBARDIER COMPLETION CENTRE INC | P-EASA.IM.A.S.03192 | | TCCA STC SA04-113 | BD-700-1A10 | Installation, complete Custom Aircraft Interior | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.CRI A-100, CRI D-100, CRI D-200, CRI D-300, CRI D-500, CRI F-100, CRIF-200The certificac (...) | Transport Canada SoC reference C-09-0069 dated 27 August 2009.-TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.-Aero Consulting Services Configuration Definition List D954000Revision DI dated 20 August 2009.-Instructions for ContinuedAirworthine (...) | Restricted to ASN 9306.1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b).3. Prior to installation of this modification the installer must determine that t (...) | 02.09.2009 |
| 10028361 | BOMBARDIER COMPLETION CENTRE INC | | | TCCA STC SA05-60 | BD-700-1A11 | Installation, Complete Custom Aircraft Interior - ASN 9353 | EASA TCDS No IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | TCCA STC No. SA05-60 Issue 1 dated 14 June 2005.-TCCA SoC ref C-09-0926 dated 15 December 2009.-Aero Consulting Services Ltd. Configuration Definition List E025000Revision CF.-Aero Consulting Services Ltd. Airplane Flight Manual SupplementG-FM25004 (...) | Restricted to ASN 9353 | 21.12.2009 |
| 10028491 | BOMBARDIER COMPLETION CENTRE INC | | | TCCA STC C-LSA09-091/D | BD-700-1A10 | Bandit Mask Paint Scheme - ASN 9332. | EASA.IM.A.009.The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.CRI D-800 - Emergency Exit Markings - Interruption of Two-InchBandOutlini (...) | Transport Canada SoC reference C-09-1183 dated 12 January 2010.TCCA LSTC No C-LSA09-091/D Issue 1 dated 10 November 2009.ACS-NAJ Ltd Modification Data Summary Number G064000 Revision N/C.*or later revisions of the above listed documents approved by (...) | Restricted to ASN 9332. | 15.01.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|---------------------------------|--|---|---|---|---|---------------|
| 10028492 | BOMBARDIER COMPLETION CENTRE INC | | | TCCA STC SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9332. | EASA.IM.A.009.The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada SoC reference C-09-0992 dated 12 January 2010.TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D954000 Revision DT dated 22 December 2009. Instructions for Continued Airworthine (...) | Restricted to ASN 9332. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.WLANS system is compatible only to IEEE 802.11(b). | 15.01.2010 |
| 10015334 | BOMBARDIER INC. | EASA.IM.A.S.00358 | | TCCA LTD STC C-LSA04-212/D | CL-600-2B16 | Installation of Securaplane Camera System Transport Canada Limited STC C-LSA04-212/D in accordance with Aero Consulting Services Modification Data Summary Notice E091000 Revision r/c or later Transport Canada Approved Revision. | | | 1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 2. If transfer of this Certificate is requested under Part 21A.116, the STC (...) | 07.12.2004 |
| 10015477 | BOMBARDIER INC. | EASA.IM.A.S.01231 | | C-LSA05-024/D | BOMBARDIER BD-700-1A10 BOMBARDIER BD-700-1A11 | Installation of Thales Integrated Electronic Standby Instruments for BD700-1A10 Global Express and BD700-1A11 Global 5000 Validation of imported TCC STC: SA04-118 | | AFM Supplement G-FM34004001 Rev B or later TCCA approved rev. | 1. ref. G-FM34004001 Rev B. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.08.2005 |
| 10015498 | BOMBARDIER INC. | EASA.IM.A.S.01261 | | C-LSA05-298/D | BOMBARDIER CL-600-2B16 (CL-604) | Aircraft Completion 11 Passenger Interior Configuration | | Aero Consulting Services Modification Data Summary E478000 revision N/C or later approved revision. Aero Consulting Services AFM Supplement E478090 revision N/C or later approved revision. | Valid only for ASN 5596 Registration OY-SGM | 07.09.2005 |
| 10015499 | BOMBARDIER INC. | EASA.IM.A.S.01262 | | TCCA STC C-LSA05-321/D | BOMBARDIER CL-600-2B16 (CL-604) | Aircraft Completion 9 Passenger Interior Configuration | | Aero Consulting Services Ltd. Modification Data Summary E264000 Revision N/C or later approved revision. Aero Consulting Services Ltd. AFM Supplement E264090 Revision N/C or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 15.11.2005 |
| 10015500 | BOMBARDIER INC. | EASA.IM.A.S.01263 | | C-LSA06-022/D | BOMBARDIER CL-600-2B16 (CL-604) | Aircraft Completion, 12 Passenger Interior Configuration | | Aero Consulting Ltd Modification Data Summary E267000 Revision N/C or later approved revision. Aero Consulting Services Ltd AFM Supplement E267090 Revision N/C or later approved revision. | Restricted to ASN 5602 | 20.01.2006 |
| 10015501 | BOMBARDIER INC. | EASA.IM.A.S.01264 | | BOMBARDIER BD-100-1A10 (CHALLE) | BOMBARDIER BD-100-1A10 (CHALLE) | Installation of an Executive interior on aircraft 20047 in accordance with MDL E449000 | | AFM Supplement E449090 | 1. This STC is limited to SN 20047.2. Certification Basis: an equivalent Safety Finding has been granted in CRI D-8 for Emergency exit Marker and Locator Signs. A deviation form paragraph 25.785(B) has been granted for personal injury criteria of dyn (...) | 07.11.2005 |
| 10015502 | BOMBARDIER INC. | EASA.IM.A.S.01265 | REV 1 | TCCA C-LSA06-051/D | BOMBARDIER BD-100-1A10 (CHALLE) | Aircraft completion, 9 passengers interior configuration in accordance with Aero Consulting Services Ltd. Modification Data Summary E840000 Revision N/C | | AFM Supplement E840090 rev N/C | 1. This STC is limited to SN 200942. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 02.08.2006 |
| 10015563 | BOMBARDIER INC. | EASA.IM.A.S.01461 | | TCCA STC Q-LSA05-030/D | BD-700-1A10 | Installation of Replacement Cockpit Printer | | Master Drawing List MDL05118003 Rev N/C Time Limits and Maintenance TL05118-004 Rev N/C | 1. Installation is limited to S/N 9086. Note: All Portable Electronic Equipment/Devices PED, must be unplugged, turned-off and stowed prior taxi, take-off and landing. 2. This STC is approved only for the product configuration as defined in the approved de (...) | 12.12.2005 |
| 10015591 | BOMBARDIER INC. | EASA.IM.A.S.01502 | | TCCA LIMITED STC C-LSA05-347/D | BOMBARDIER BD-700-1A10 | Installation, Aircraft Interior Completion for BD700-1A10 Global Express acc.: MDS E583000 Rev. N/C Validation of imported TCCA Limited STC: C-LSA05-347/D | | AFM Supplement G-FM25000133 Rev. N/C or later TCCA approved rev. ICA GC91630001-IFCAAMM Supplement/Airworthiness Limitations G-TM05100069 | ref. G-FM25000133 Rev. N/C. Limited to S/N 9163. Note: 1. The airplane is limited to private use only unless the mid cabin door is rendered inoperable. 2. Approval of EFB installation does not constitute operational approval to use EFB function. 3. Th (...) | 17.11.2005 |
| 10015594 | BOMBARDIER INC. | EASA.IM.A.S.01507 | | FAA STC SA556NE | CL-600-2B16 | Installation of a Forward Overboard Servicing | | LA E.L. Modification List ML 13545 Revision A dated January 7 1988 or later approved revision | Limited to Airplane Serial Number 5055 | 25.10.2005 |
| 10015630 | BOMBARDIER INC. | EASA.IM.A.S.01582 | REV. 1 | TCCA STC SA04-113 | BD-700-1A10 | Installation, Completion of custom Aircraft Interior for BD700-1A10 Global Express acc.: CDL D954000 Rev. K Validation of imported TCC STC: SA04-113 | | AFM Supplement G-FM25004001 Rev N/C or later TCCA approved rev. ICA GC91450000-IFCAAMM Supplement G-TM05104001 | 1. Ref. G-FM25004001 Rev. N/C. Limited to S/N 91452. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 12.09.2005 |
| 10015641 | BOMBARDIER INC. | EASA.IM.A.S.01605 | | TCCA STC C-LSA06-014/D | BOMBARDIER BD-700-1A11 | Installation 13 Passenger interior completion for BD700-1A11 | | Master Drawing List E660000 Rev N/C-Flight Manual Supplement G-FM25000135 Rev N/C-Time Limits and Maintenance GC91680001-TLMC Rev N/C | 1. Installation of limited to S/N 91682. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 30.01.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------|-------------------|----------------|----------------------------|---------------------------------|--|---|---|--|---------------|
| 10015642 | BOMBARDIER INC. | EASA.IM.A.S.01606 | REV. 1 | TCCA STC C-LSA05-391/D | BD-700-1A10 | Aircraft Completion, 13 Passenger Interior Configuration | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | EASA STC No. EASA.IM.A.S.01606-Transport Canada SoC ref.C-10-0444 dated 25 June 2010- Transport Canada STC No. C-LSA05-391/D Issue 1 dated 22 December 2005- Aero Consulting Services Master Drawing List E659000 Revision D dated 3 June 2010- Flight (...) | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. 2. The installation is limited to S/N 9169 | 07.07.2010 |
| 10015643 | BOMBARDIER INC. | EASA.IM.A.S.01607 | | TCCA STC NO C-LSA05-382/D | BOMBARDIER BD-700-1A11 | Aircraft Completion 13 Passenger Interior Configuration/Validation of TCCA Limited STC C-LSA05-382/D to S/N 9170MDL: E810000 Rev A or later approved revisions | | FMS: G-FM25000136 Rev A or later approved revisions/IFCA: GC91700001-IFCA | | 07.04.2006 |
| 10015675 | BOMBARDIER INC. | EASA.IM.A.S.01656 | | FAA STC ST03065AT | LEARJET 31A | Installation of a Single Universal UNS-1C+ or UNS-1E Flight Management System | | Bombardier Aerospace Master Document List BAS31017000 Revision B dated February 22 2005 or later approved revision/Bombardier Aerospace AFM Supplement ref. BAS31047000 Revision NC dated March 14 2005 or later approved revision. | | 04.01.2006 |
| 10015737 | BOMBARDIER INC. | EASA.IM.A.S.01761 | | | BOMBARDIER BD-700-1A10 | Customised Interior Completion Aircraft 20071, in accordance with Aero Consulting Services Ltd. Modification Data Summary E725000 rev N/C or later approved revision | | AFM Supplement E725090 rev N/N or later approved revision | 1. S/N 20071 | 30.01.2006 |
| 10015743 | BOMBARDIER INC. | EASA.IM.A.S.01767 | | FAA STC ST00393DE-D | LEARJET 31A | Installation of a Honeywell 440, 660 or 880 Radar System | | 1) Learjet Inc. Master Document List BAS31010014 Revision B dated July 06, 2000 or later approved revision. 2) AFM Supplement BAS31040109 Revision (-) dated October 06, 1999 or later approved revision. | None | 20.01.2006 |
| 10015781 | BOMBARDIER INC. | EASA.IM.A.S.01827 | | TCCA LTD STC C-LSA06-132/D | BOMBARDIER BD-700-1A10 | Customized Interior Completion ACSN 9173/Validation of TCCA Limited STC C-LSA06-132/D to S/N 9173MDL: E776000 Rev. NC or later TCCA approved revisions/TCCA SoC: C-06-316 dated June 30, 2006 | | FMS: G-FM25000138 Rev. NC or later TCCA approved revisions/IFCA: GC91730000-IFCA | Aerworthiness Limitations - Camera System Electrical Receptacle Installation - F806: G-TM05100073 Rev. NC or later TCCA approved revisions | 17.07.2006 |
| 10015841 | BOMBARDIER INC. | EASA.IM.A.S.01936 | | | BOMBARDIER BD-100-1A10 (CHALLE) | Aircraft completion, interior configuration in accordance with Aero Consulting Services Ltd. Modification Data Summary E843000 revision N/C | | AFM Supplement E843090 rev N/C | Limited to S/N 20097 | 17.08.2006 |
| 10015858 | BOMBARDIER INC. | EASA.IM.A.S.01962 | | TCCA STC C-LSA06-020/D | BOMBARDIER BD-100-1A10 | Aircraft completion, 10 Passenger interior configuration, in accordance with MDS E790000 | | AFM Supplement E790090 | S/N 20079 | 24.04.2006 |
| 10015903 | BOMBARDIER INC. | EASA.IM.A.S.02029 | | TCCA STC Q-LSA06-25/D | BOMBARDIER BD-700-1A10 | Installation Acoustic Insulation ACSN 9163/Validation of TCCA Limited STC Q-LSA06-25/D to S/N 9163MDL: 26118002 Rev NC or later TCCA approved revisions/TCCA SoC: NAI5012-06-00062 dated June 20, 2006 | | -- | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.07.2006 |
| 10015912 | BOMBARDIER INC. | EASA.IM.A.S.02041 | | C-LSA06-270/D | BOMBARDIER BD-100-1A10 (CHALLE) | Aircraft completion, 10 passenger interior configuration, in accordance with MDS F029000 (Original STC number: Transport/Canada STC C-LSA06-270/D) | | AFM Supplement F029090 | Limited to S/N 20106 | 18.10.2006 |
| 10015956 | BOMBARDIER INC. | EASA.IM.A.S.02099 | REV. 1 | TCCA STC SA06-37 | BD-700-1A10, BD-700-1A11 | Installation, Provisions for D(-&->)M PN 444-DMC57-1 Securaplane/Antenna. | EASA IM.A.009.The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | EASA STC No. EASA.IM.A.S.02099 issued 30 November 2006.Transport Canada STC No. SA06-37Transport Canada SoC ref. C-05 0510 dated 23 November 2006.Aero Consulting Services MDL No. E448000 Revision A.Aero Consulting Services MMMS G-TM05104501 or G-TM0 (...) | None. | 17.03.2010 |
| 10015997 | BOMBARDIER INC. | EASA.IM.A.S.02154 | | TCCA STC C-LSA07-003/D | BOMBARDIER BD-100-1A10 (CHALLE) | Aircraft completion, 10 passenger interior configuration, in accordance with MDS f070000 (Original STC number: Transport Canada STCC-LSA07-003/D) | | AFM Supplement F070090. | Limited to S/N 20111. | 12.01.2007 |
| 10016019 | BOMBARDIER INC. | EASA.IM.A.S.02186 | REV. 63 | TCCA STC SA06-90 | CL-600-2B16 (604 VARIANT) | Installation of Complete Customized Aircraft Interior | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Aero Consulting Services Ltd. Configuration Definition List Ref.E934000 Rev.HL dated 07, December 2012.- Bombardier Airplane Flight Manual Supplement EFM-2500-4000 Revision B.- Aero Consulting Services Ltd. Airplane Flight Manual Supplement AFMSE97 (...) | EASA approved configurations are identified in the Configuration/Definition List document Ref. E934000.Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed (...) | 21.03.2013 |
| 10016102 | BOMBARDIER INC. | EASA.IM.A.S.02292 | | TCCA C-LSA07-049/D ISSUE 1 | BOMBARDIER BD-100-1A10 (CHALLE) | AIRCRAFT COMPLETION, 10 PASSENGER INTERIOR CONFIGURATION/EASA Validation of:Transport Canada Limited Supplemental Type Certificate C-LSA07-049/D Issue 1 dated 19th February 2007, in accordance with Aero Consulting Services Ltd. Modification Data Summary F075000 Revision | | Transport Canada Limited Supplemental Type Certificate C-LSA07-049/D Issue 1 dated 19th February 2007/Transport Canada Approved Aero Consulting Services Ltd Airplane Flight Manual Supplement F075090 Revision N/C.Instructions for Continued Airworthiness HC201165000-IFCA-1 | 1. Compatibility with previous approved modifications shall be determined by the installer. 2. This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20116. | 28.02.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------|-------------------|----------------|--------------------------------|---|--|--|---|--|---------------|
| 10016120 | BOMBARDIER INC. | EASA.IM.A.S.02312 | | TCCA STC NO. C-LSA07-154/D | BOMBARDIER BD-100-1A10 (CHALLE) | AIRCRAFT COMPLETION, 10 PASSENGER INTERIOR CONFIGURATION /EASA Validation of:Transport Canada Limited Supplemental TypeCertificate C-LSA07-154/Issue 1 dated 15th June 2007, in accordance with Aero Consulting Services Ltd. Modification Data Summary F311000 Revision N/C. | | # Transport Canada Limited Supplemental Type Certificate C-LSA07-154/D Issue 1, dated 15th June 2007.# Transport Canada Approved Aero Consulting Services Ltd. Airplane Flight Manual Supplement F311090 Revision N/C.# Instructions for Continued Airworthiness HC201455000-IFCA-1 | 1. This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20145.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modi (...) | 03.07.2007 |
| 10016197 | BOMBARDIER INC. | EASA.IM.A.S.02437 | | TCCA STC C-LSA07-288/D | BOMBARDIER BD-100-1A10 (CHALLE) | Aircraft Completion, 10 passengers interior configuration:EASA Validation of:Transport Canada Limited Supplemental TypeCertificate C-LSA07-288/D, Issue 1 dated 22nd October 2007, in accordance with Aero Consulting Services Ltd. Modification Data Summary F494000 Revision A dated 25.10.2007. | | Transport Canada Limited Supplemental Type Certificate C-LSA07-288/D Issue 1, dated 22nd October 2007.Transport Canada Approved Aero Consulting Services Ltd. Airplane Flight Manual Supplement F494090 Revision N/C or later Transport Canada approved r (...) | This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20155. | 31.10.2007 |
| 10016239 | BOMBARDIER INC. | EASA.IM.A.S.02506 | REV. 4 | TCCA STC SA07-117 | BOMBARDIER BD-700-1A10 BOMBARDIER BD-700-1A11 | Installation of Class 2 Electronic Flight Bag (EFB). | EASA TCDS No. IM.A.009The Certification Basis for the original product remains applicable to this certificate.The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate. | EASA STC No. EASA.IM.A.S.02508 (10016239) Rev 2 dated 24 September 2009.TCCA STC No. SA07-117 Issue 6 dated 30 July 2009.TCCA SoC reference C-10-0345 dated 04 May 2010.Aero Consulting Services Modification Data Summary Number F376000Revision M dated (...) | This installation approval does not constitute operational approval for any use of the EFB. This approval is for the EFB system structural mounting and interconnect wiring only. | 17.02.2011 |
| 10016251 | BOMBARDIER INC. | EASA.IM.A.S.02528 | | TCCA STC C-LSA08-012/D | BOMBARDIER BD-100-1A10 (CHALLE) | AIRCRAFT COMPLETION, 10 PASSENGERS INTERIOR CONFIGURATION /EASA Validation of:Transport Canada Limited Supplemental Type Certificate C-LSA08-012/D Issue 1 dated 11th January 2008, in accordance with Aero Consulting Services Ltd. Modification Data Summary F579000 | | Transport Canada Limited Supplemental Type Certificate C-LSA08-012/D Issue 1, dated 11th January 2008. Transport Canada Approved Aero Consulting Services Ltd. Modification Data Summary ref. F579000 dated 22nd January 2008, Revision A o (...) | 1. This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20172.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modi (...) | 25.01.2008 |
| 10016293 | BOMBARDIER INC. | EASA.IM.A.S.02589 | | TCCA STC SA06-84 | BOMBARDIER BD-100-1A10 (CHALLE) | Improved Hinge Design, Toilet Seat and Lid / Modification of existing toilet seat installation which allows both toilet seat and lid to remain in the up position without falling.EASA Validation of Transport Canada Supplemental Type Certificate SA06-8 (...) | | -Transport Canada Supplemental Type Certificate SA06-84 Issue 1, dated 3rd November 2006.-Aero Consulting Modification Data Summary, ref. No. F062000 Revision A, dated 02.08.2007. | -Decrease Aircraft Service Bulletin "Bombardier C300 Lavatory Seat Lid/Hinge Assembly", ref. SB06008, Revision G, dated 23rd July 2007. | 05.11.2007 |
| 10016304 | BOMBARDIER INC. | EASA.IM.A.S.02601 | REV. 1 | TCCA STC C-LSA07-285/D | BD-700-1A10 | Aircraft Completion, 14 Passenger Interior Configuration. | EASA IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | EASA STC No. EASA.IM.A.S.02601.TCCA STC No. C-LSA07-285/D Issue 1 dated 02 November 2007.TCCA SoC ref. C-10-0426 dated 01 June 2010.Aero Consulting Services Modification Data Summary F461000 Revision Adated 27 April 2010.Aero Consulting Services File (...) | Restricted to ASN 9237.This does not constitute an operational approval of the Wireless Local Area Network (WLAN) system. It is the operator's responsibility to obtain operational approval for the airborne use of transmitting portable electronic device (...) | 07.06.2010 |
| 10016332 | BOMBARDIER INC. | EASA.IM.A.S.02641 | REV. 1 | TCCA STC C-LSA07-229/D | BD-700-1A11 | Installation, Interior Completions In-Service Improvement Modifications for ASN 9249. | EASA IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged (...) | EASA STC no. EASA.IM.A.S.02641-TCCA STC no. C-LSA07-229/D Issue 1 dated 19 December 2007-ACS-NAI Ltd. Modification Data Summary F428000 Revision B or later approved revision.-Bombardier Airplane Flight Manual Supplement G-FM25000163 Revision N/C.-In (...) | Restricted to ASN 9249 | 20.04.2011 |
| 10016333 | BOMBARDIER INC. | EASA.IM.A.S.02642 | | TCCA STC NO. SA04-112 ISSUE 3. | BOMBARDIER BD-100-1A10 (CHALLE) | AIRCRAFT COMPLETION, 10 PASSENGERS INTERIOR CONFIGURATION /EASA Validation of:Transport Canada Limited Supplemental Type Certificate C-LSA07-411/D Issue 1 dated 5th December 2007, in accordance with Aero Consulting Services Ltd. Modification Data Summary F637000 Revision N/C dated 05.12.2007 | | Transport Canada Limited Supplemental Type Certificate C-LSA07-411/D Issue 1, dated 5th December 2007. Transport Canada Approved Aero Consulting Services Ltd. Modification Data Summary F637000, Revision N/C or later Transport Canada appr (...) | This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20141. | 18.12.2007 |
| 10016427 | BOMBARDIER INC. | EASA.IM.A.S.02766 | | TCCA STC C-LSA08-145/D ISSUE 1 | BOMBARDIER BD-100-1A10 (CHALLE) | CUSTOMIZED INTERIOR COMPLETION # AIRCRAFT 2020310 passengers seats interior configuration:EASA Validation of:Transport Canada Limited Supplemental Type Certificate C-LSA08-145/D Issue 1 dated 18th August 2008, in accordance with Aero Consulting Services (...) | | Transport Canada Limited Supplemental Type Certificate C-LSA08-145/D Issue 1, dated 18th August 2008. Transport Canada Approved Aero Consulting Services Ltd. Modification Data Summary ref. F827000 dated 28th July 2008, Revision N/C or la (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.1 (...) | 20.08.2008 |
| 10016446 | BOMBARDIER INC. | EASA.IM.A.S.02792 | | STC C-LSA08-107/D | BOMBARDIER BD-100-1A10 (CHALLE) | AIRCRAFT COMPLETION, 10 PASSENGERS INTERIOR CONFIGURATION /EASA Validation of:Transport Canada Limited Supplemental Type Certificate C-LSA08-107/D Issue 1 dated 19th March 2008, in accordance with Aero Consulting Services Ltd. Modification Data Summary F786000 Revision N/C. | | Transport Canada Limited Supplemental Type Certificate C-LSA08-107/D Issue 1, dated 19th March 2008. Transport Canada Approved Aero Consulting Services Ltd. Modification Data Summary ref. F786000 dated 19th March 2008, Revision N/C or (...) | This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20123. | 28.03.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------|-------------------------|----------------|-----------------------|--------------------------|--|--|---|--|---------------|
| 10016478 | BOMBARDIER INC. | P- EASA.IM.A.S.02830 | REV. 4 | TCCA STC SA04-113 | BD-700-1A10 | Aircraft Completion, 13 Passenger Interior Configuration | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -EASA STC No. IM.A.S.02830 (10016478) Revision 2 dated 17 June 2010. -TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. -Aero Consulting Services Configuration Definition List D954000 Revision1G dated 18 February 2011. -Instructions for Continued Airworthiness (...) | Restricted to ASN 9274 and 1) Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2) WLAN system is compatible only to IEEE 802.11(b). | 12.04.2011 |
| 10016544 | BOMBARDIER INC. | EASA.IM.A.S.02923 | REV. 5 | TCCA STC SA08-48 | BD-700-1A10, BD-700-1A11 | Installation, Interior Completions In-Service Improvement Modifications | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged (...) | TCCA STC ref. C-11-0232 dated 27 April 2011. TCCA STC No. SA08-48 Issue 2 dated 27 August 2009. ACS-NAI Ltd. Modification Data Summary Number F727000 Revision H or later approved revision or later revisions of the above listed documents approved by EASA in (...) | None | 11.05.2011 |
| 10016660 | BOMBARDIER INC. | EASA.IM.A.S.03121 | REV. 1 | TCCA STC SA05-60 | BD-700-1A11 | Installation, Complete Custom Aircraft Interior (S/N 9311). | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | EASA STC No. EASA.IM.A.S.03121. TCCA STC No. SA05-60 Issue 1. Aero Consulting Services Ltd Configuration Definition List No. E025000 Revision CT dated 31 May 2010. or later revisions of the above listed documents approved by EASA in accordance with EASA (...) | Restricted to ASN 9311. | 04.06.2010 |
| 10027337 | BOMBARDIER INC. | | | TCCA STC SA04-113 | BOMBARDIER BD-700-1A10 | Installation, Complete Custom Aircraft Interior | EASA.IM.A.009The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate. CRI A-100, CRI D-100, CRI D-200, CRI D-300, CRI D-500, CRI F-100, CRIF-200. | Transport Canada SoC reference C-09-0070 dated 14 September 2009. TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D954000 Revision DK dated 08 September 2009. Instructions for Continued Airworthiness (...) | Restricted to ASN 9309. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS. WLAN system is compatible only to IEEE 802.11(b) | 22.09.2009 |
| 10027762 | BOMBARDIER INC. | | REV. 1 | TCCA STC SA05-60 | BD-700-1A11 | Installation, Complete Custom Aircraft Interior - ASN 9343 | EASA TCDS No IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. SA05-60 Issue 1, dated 14 June 2005. Aero Consulting Services Ltd Configuration Definition List No. E025000 Revision DB dated 31 May 2010. Bombardier's Airplane Flight Manual Supplement G-FM25004501 Revision N/C. GC93430000-IFCA Instructions (...) | Restricted to ASN 9343. | 04.06.2010 |
| 10028181 | BOMBARDIER INC. | | | TCCA STC SA05-60 | BD-700-1A11 | Installation, Complete Custom Aircraft Interior - ASN 9231 | EASA TCDS No IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No SA05-60 Issue 1 dated 14 June 2005. TCCA SoC ref C-09-1021 dated 2 December 2009. Aero Consulting Services Ltd. Configuration Definition List E025000 Revision CC. Aero Consulting Services Ltd. Airplane Flight Manual Supplement G-FM25004501 Rev (...) | Restricted to ASN 9231 | 07.12.2009 |
| 10028344 | BOMBARDIER INC. | | | | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9320 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada SoC reference C-09-0275 dated 09 December 2009. TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D954000 Revision DQ dated 08 December 2009. Instructions for Continued Airworthiness (...) | Restricted to ASN 9320 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2) WLAN system is compatible only to IEEE 802.11(b). 3. Prior to installation of this modification the installer must determine the (...) | 18.12.2009 |
| 10028416 | BOMBARDIER INC. | | REV. 2 | TCCA STC SA04-113 | BD-700-1A10 | Modification to install the Fwd Pocket door and Mid Pocket door on a separate circuit. Installation, Complete Custom Aircraft Interior - Revision to EASA Supplemental Type Certificate 10028416 to incorporate missing drawings on the MDL - ASN 9329 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -EASA STC No. 10028416 Rev. 1 dated 3 February 2010. -Transport Canada SoC reference C-10-0158 dated 17 May 2010. -TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. -Aero Consulting Services Configuration Definition List D954000 Revision EG dated 09 A (...) | Restricted to ASN 9329 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2) WLAN system is compatible only to IEEE 802.11(b). | 19.05.2010 |
| 10028519 | BOMBARDIER INC. | | | TCCA STC NO. SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9333 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. Non-significant. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada SoC reference C-09-0993 dated 14 January 2010. TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D954000 Revision DU dated 23 December 2009. Instructions for Continued Airworthiness (...) | Restricted to ASN 9333 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2) WLAN system is compatible only to IEEE 802.11(b). | 21.01.2010 |
| 10028553 | BOMBARDIER INC. | | | TCCA STC NO. SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9115 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. Non-significant. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada SoC reference C-09-0721 dated 14 January 2010. TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D954000 Revision DG dated 29 July 2009. Instructions for Continued Airworthiness G (...) | Restricted to ASN 9115 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2) WLAN system is compatible only to IEEE 802.11(b). | 21.01.2010 |
| 10028579 | BOMBARDIER INC. | | REV. 2 | TCCA STC SA05-60 | BD-700-1A11 | Installation, Interior Completions In-Service Improvement Modifications - ASN 9334 | EASA TCDS No IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged (...) | TCCA STC No SA05-60 Issue 1 dated 14 June 2005. Aero Consulting Services Ltd. Configuration Definition List E025000 Revision EW dated 22 June 2011. Bombardier Airplane Flight Manual Supplement G-FM25004501 Revision N/Cor later approved revision GC93340 (...) | Restricted to ASN 9334 | 08.08.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------|-------------------|----------------|------------------------|-------------------|---|---|---|--|---------------|
| 10029275 | BOMBARDIER INC. | | | TCCA STC SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9323. | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada SoC reference C-10-0011 dated 11 March 2010.TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.Aero Consulting Services Configuration Definition List D954000 RevisionDY dated 03 March 2010.Instructions for Continued Airworthiness G (...) | Restricted to ASN 9323. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.WLAN system is compatible only to IEEE 802.11(b). | 17.03.2010 |
| 10029589 | BOMBARDIER INC. | | | TCCA STC SA10-19 | BD-700-1A10 | Installation, Upgrade from Swift to Swift Broadband | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. SA10-19 Issue 1 dated 08 February 2010.-TCAA SoC ref. C-10-0232 dated 24 March 2010.-ACS-NAI Ltd. Modification Data Summary Number G541000 Revision NC or later approved revision.-Instructions for Continued Airworthiness G541071 Revision (...) | None | 02.04.2010 |
| 10029628 | BOMBARDIER INC. | | | TCCA STC SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9336 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada SoC reference C-10-0275 dated 07 April 2010.-TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.-Aero Consulting Services Configuration Definition List D954000 RevisionED dated 31 March 2010.-Instructions for Continued Airworthiness (...) | Restricted to ASN 9336 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2. WLAN system is compatible only to IEEE 802.11(b). | 08.04.2010 |
| 10029774 | BOMBARDIER INC. | | | TCCA STC SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9341 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada SoC reference C-10-0151 dated 08 April 2010.-TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.-Aero Consulting Services Configuration Definition List D954000 RevisionEE dated 31 March 2010.-Instructions for Continued Airworthiness (...) | Restricted to ASN 9341 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2. WLAN system is compatible only to IEEE 802.11(b). | 20.04.2010 |
| 10030001 | BOMBARDIER INC. | | | TCCA STC SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9339 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada SoC reference C-10-0317 dated 06 May 2010.-TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.-Aero Consulting Services Configuration Definition List D954000 RevisionEU dated 23 April 2010.-Instructions for Continued Airworthiness (...) | Restricted to ASN 9339 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2. WLAN system is compatible only to IEEE 802.11(b). | 10.05.2010 |
| 10030040 | BOMBARDIER INC. | | | TCCA STC NO. SA10-15 | BD-700-1A10/-1A11 | Modification, External Airstair Door Switch | EASA.IM.A.009The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA SoC ref. C-10-0116 dated 23 April 2010.TCCA STC No. SA10-15 Issue 1 dated 02 February 2010.ACS-NAI Ltd. Modification Data Summary Number G572000 Revision NC or later approved revision.ACS-NAI Ltd Instructions for Continued Airworthiness G572071 (...) | None | 17.05.2010 |
| 10030078 | BOMBARDIER INC. | | | TCCA STC C-LSA02-396/D | BD-700-1A10 | Modification, 13 Passenger Interior Configuration | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. C-LSA02-396/D Issue 1 dated 14 November 2002.TCCA SoC ref. C-10-0369 dated 17 May 2010.Aero Consulting Services (ACS) Modification Data Summary D12050 RevisionN/C or later approved revision.ACS Flight Manual Supplement *Conversion: 13 P (...) | Restricted to ASN 9102.1. Previous Embodiment of EASA STC No. 10030079 "Aircraft Completion, 10 Passenger Interior Configuration" is required.2. Bombardier Service Bulletin 700-31-018 Modification - Integrated Avionics Computers - Avionics 2001 Batch 1 IAC Upgrade must be embodied. | 19.05.2010 |
| 10030079 | BOMBARDIER INC. | | | TCCA STC C-LSA02-330/D | BD-700-1A10 | Aircraft Completion, 10 Passenger Interior Configuration | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. C-LSA02-330/D Issue 1 dated 24 October 2002.-TCCA SoC ref. C-10-0368 dated 14 May 2010.-Aero Consulting Services (ACS) Modification Data Summary D12000 Revision C or later approved revision. The following ACS Flight Manual Supplements are (...) | Restricted to ASN 9102 | 19.05.2010 |
| 10030080 | BOMBARDIER INC. | | | TCCA STC SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9331 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada SoC reference C-09-0994 dated 13 May 2010.-TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.-Aero Consulting Services Configuration Definition List D954000 RevisionEI dated 16 April 2010.-Instructions for Continued Airworthiness (...) | Restricted to ASN 9331 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2. WLAN system is compatible only to IEEE 802.11(b). | 19.05.2010 |
| 10030239 | BOMBARDIER INC. | EASA.IM.A.S.03164 | REV. 1 | TCCA STC SA05-60 | BD-700-1A11 | Installation, Interior Completions. | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | EASA STC No. EASA.IM.A.S.03164.TCCA STC No. SA05-60 Issue 1.Aero Consulting Services Ltd Configuration Definition List No. E025000 Revision CV dated 31 May 2010. or later revisions of the above listed documents approved by EASA in accordance with EASA (...) | Restricted to ASN 9318. | 04.06.2010 |
| 10030240 | BOMBARDIER INC. | EASA.IM.A.S.03162 | | TCCA STC SA05-60 | BD-700-1A11 | Installation, Interior Completions. | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | EASA STC No. EASA.IM.A.S.03162.TCCA STC No. SA05-60 Issue 1.Aero Consulting Services Ltd Configuration Definition List No. E025000 Revision CU dated 31 May 2010. or later revisions of the above listed documents approved by EASA in accordance with EASA (...) | Restricted to ASN 9315. | 04.06.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------|-------------------|----------------|--------------------------|-------------|---|---|--|---|---------------|
| 10030781 | BOMBARDIER INC. | | REV. 1 | TCCA STC SA04-113 | BD-700-1A10 | Installation, Interior Completions In-Service Improvements Modifications- ASN 9348 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | Transport Canada SoC reference C-11-0065 dated 17 February 2011. EASA STC No. 10030781. TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D954000 Revision GD dated 01 February 2011. Instructions for Continued Airworthiness (...) | Restricted to ASN 9348 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b). | 25.02.2011 |
| 10030900 | BOMBARDIER INC. | | REV. 3 | TCCA STC SA05-60 | BD-700-1A11 | Installation, Interior Completions In-Service Improvements Modifications- ASN 9359 | EASA TCDS No. IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | TCCA STC No. SA05-60 Issue 1 dated 14 June 2005. EASA STC No. 10030900 Rev. 1. TCAA SoC ref. C-11-0062 dated 16 February 2011. Aero Consulting Services Ltd. Configuration Definition List E025000 Revision EK dated 1 February 2011. Bombardier Aerospace Airplane Flight Manual Supplement G (...) | Restricted to ASN 9359 | 02.03.2011 |
| 10031088 | BOMBARDIER INC. | | REV. 1 | TCCA STC SA05-60 | BD-700-1A11 | Installation, Interior Completions In-Service Improvements Modifications- ASN 9373 | EASA TCDS No. IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | TCCA STC No. SA05-60 Issue 1 dated 14 June 2005. TCCA SoC ref. C-11-0063 dated 16 February 2011. EASA STC No. 10031088. Aero Consulting Services Ltd. Configuration Definition List E025000 Revision EL dated 01 February 2011. Bombardier Aerospace Airplane Flight Manual Supplement G (...) | Restricted to ASN 9373 | 24.02.2011 |
| 10031105 | BOMBARDIER INC. | EASA.IM.A.S.03036 | REV. 1 | TCCA STC SA 08-65 | BD700-1A10 | Water Tank Modification | EASA TCDS No. EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | EASA STC No. EASA.IM.A.S.03036 dated 02 June 2009. TCCA STC No. SA08-65 Issue 3 dated 02 July 2010. TCCA SoC reference C-10-0260 dated 21 July 2010. Aero Consulting Services Modification Data Summary Number F250000 Revision C or later approved revision (...) | None | 27.07.2010 |
| 10031184 | BOMBARDIER INC. | | | TCCA STC SA04-113 | | Installation, Complete Custom Aircraft Interior - ASN 9351 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | Transport Canada SoC reference C-10-0355 dated 29 July 2010. TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D954000 Revision FN dated 27 July 2010. Instructions for Continued Airworthiness GC (...) | Restricted to ASN 9351 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b). 3. Prior to installation of this modification the installer must determine that (...) | 30.07.2010 |
| 10031510 | BOMBARDIER INC. | | | TCCA STC SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9349 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | Transport Canada SoC reference C-10-0308 dated 21 July 2010. TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D954000 Revision FH dated 09 July 2010 or later revisions of the above listed documents (...) | Restricted to ASN 9349. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b). | 20.08.2010 |
| 10031663 | BOMBARDIER INC. | | | TCCA STC SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9364 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | Transport Canada SoC reference C-10-0310 dated 30 August 2010. TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D954000 Revision FO dated 18 August 2010. Instructions for Continued Airworthiness (...) | Restricted to ASN 9364 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b). 3. Prior to installation of this modification the installer must determine that (...) | 02.09.2010 |
| 10031900 | BOMBARDIER INC. | | | TCCA STC SA05-60 ISSUE 1 | BD-700-1A11 | Installation, Complete Custom Aircraft Interior - ASN 9379 | EASA TCDS No. IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | TCCA STC No. SA05-60 Issue 1 dated 14 June 2005. TCCA SoC ref. C-10-0356 dated 17 September 2010. Aero Consulting Services Ltd. Configuration Definition List E025000 Revision DR dated 30 July 2010. Bombardier Aerospace Airplane Flight Manual Supplement G (...) | Restricted to ASN 9379 | 20.09.2010 |
| 10032248 | BOMBARDIER INC. | | REV. 1 | TCCA STC NO. SA05-60 | BD-700-1A11 | Modification to incorporate changes to the installation of Custom Aircraft Interior - ASN 9370. Revision to EASA STC 10032248 | EASA TCDS No. IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | TCCA STC No. SA05-60 Issue 1 dated 14 June 2005. TCAA SoC ref. C-10-0926 dated 1 November 2010. Aero Consulting Services Ltd. Configuration Definition List E025000 Revision DY dated 26 October 2010. Bombardier Aerospace Airplane Flight Manual Supplement G (...) | Restricted to ASN 9370 | 09.11.2010 |
| 10032327 | BOMBARDIER INC. | | | TCCA STC NO. SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9360 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | Transport Canada SoC reference C-10-0424 dated 21 October 2010. TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D954000 Revision FS dated 18 October 2010. Instructions for Continued Airworthiness (...) | Restricted to ASN 9360 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b). 3. Prior to installation of this modification the installer must determine that (...) | 27.10.2010 |
| 10032532 | BOMBARDIER INC. | | | TCCA STC NO. SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9377 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. Non-significant. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | Transport Canada SoC reference C-10-0644 dated 10 November 2010. TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D954000 Revision FT dated 04 November 2010. Instructions for Continued Airworthiness (...) | Restricted to ASN 9377 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b). | 12.11.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------|---------------------|----------------|-------------------------------|-------------|---|---|---|---|---------------|
| 10032907 | BOMBARDIER INC. | | | TCCA STC NO. SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9367 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product areunch (...) | Transport Canada SoC reference C-10-0643 dated 02 December 2010TCCA STC No. SA04-113 Issue 1 dated 28 October 2004Aero Consulting Services Configuration Definition List D954000 RevisionFW dated 26 November 2010Instructions for Continued Airworthiness (...) | Restricted to ASN 9367 and1. Refer to the Limitations sections of the required AFM Supplement andto the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b).3.Prior to installation of this modification the installer mustdetermine that (...) | 07.12.2010 |
| 10032919 | BOMBARDIER INC. | | | TCCA STC NO. SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9371 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product areunch (...) | Transport Canada SoC reference C-10-0641 dated 25 November 2010TCCA STC No. SA04-113 Issue 1 dated 28 October 2004Aero Consulting Services Configuration Definition List D954000 RevisionFU dated 19 November 2010Instructions for Continued Airworthiness (...) | Restricted to ASN 9371 and1. Refer to the Limitations sections of the required AFM Supplement andto the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b).3.Prior to installation of this modification the installer mustdetermine that (...) | 08.12.2010 |
| 10033177 | BOMBARDIER INC. | | | TCCA STC NO. SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9369 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product areunch (...) | Transport Canada SoC reference C-10-1060 dated 16 December 2010TCCA STC No. SA04-113 Issue 1 dated 28 October 2004Aero Consulting Services Configuration Definition List D954000 RevisionFX dated 14 December 2010Instructions for Continued Airworthiness (...) | Restricted to ASN 9369 and1. Refer to the Limitations sections of the required AFM Supplement andto the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b). | 21.12.2010 |
| 10033710 | BOMBARDIER INC. | | | TCCA STC NO. SA04-113 ISSUE 1 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9385 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product areuncha (...) | Transport Canada SoC reference C-10-0758 dated 02 February 2011.TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.Aero Consulting Services Configuration Definition List D954000 RevisionGB dated 28 January 2011.Instructions for Continued Airworthin (...) | Restricted to ASN 9385 and1. Refer to the Limitations sections of the required AFM Supplement andto the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b). | 04.02.2011 |
| 10033779 | BOMBARDIER INC. | P-EASA.IM.A.S.03145 | | TCCA STC SA08-86 | BD-700-1A10 | Ground Service Mode DC Cabin Power Distribution Modification | EASA TCDS IM.A.009 | -TCCA STC No. SA08-86 Issue 1 dated 08 December 2008-TCCA SoC reference C-09-0423 dated 24 June 2009Aero Consulting Services Modification Data Summary No F928000 RevisionN/C or later approved revision-BAS67050010-ICA Instructions for Continued Airworthiness | None | 23.07.2009 |
| 10033995 | BOMBARDIER INC. | | | TCCA STC SA04-113, ISSUE 1 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9380. | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product areunch (...) | Transport Canada SoC reference C-10-0842 dated 17 February 2011.TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.Aero Consulting Services Configuration Definition List D954000 RevisionGE dated 09 February 2011.Instructions for ContinuedAirworthin (...) | Restricted to ASN 9380 and1. Refer to the Limitations sections of the required AFM Supplement andto the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b). | 24.02.2011 |
| 10033998 | BOMBARDIER INC. | | REV. 1 | TCCA STC SA05-60 | BD-700-1A11 | Aft Lavatory Backsplash replacementInstallation, Complete Custom Aircraft Interior - ASN 9405 | EASA TCDS No IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product (...) | TCCA STC No. SA05-60 Issue 1 dated 14 June 2005.Aero Consulting Services Ltd Configuration Definition List E025000Revision EU dated 03 June 2011.Bombardier Aerospace Aeroplane Flight Manual Supplement G-FM25004501Revision N/C or later approved revis (...) | Restricted to ASN 9405. | 29.06.2011 |
| 10034082 | BOMBARDIER INC. | | | TCCA STC NO. SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9394 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product areuncha (...) | TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.Aero Consulting Services Configuration Definition List D954000 RevisionGI dated 01 March 2011.Instructions for Continued Airworthiness GC93940000-IFCA or later revisions of the above listed documen (...) | Restricted to ASN 9394 and1. Refer to the Limitations sections of the required AFM Supplement andto the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b). | 04.03.2011 |
| 10034192 | BOMBARDIER INC. | | | TCCA STC NO. SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9391 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product areuncha (...) | TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.Aero Consulting Services Configuration Definition List D954000 RevisionGIH dated 22 February 2011.Instructions for Continued Airworthiness GC93910000-IFCA or later revisions of the above listed docu (...) | Restricted to ASN 9391 and1. Refer to the Limitations sections of the required AFM Supplement andto the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b). | 15.03.2011 |
| 10034584 | BOMBARDIER INC. | | | TCCA STC NO. SA05-60 | BD-700-1A11 | Installation, Complete Custom Aircraft Interior - ASN 9408 | EASA TCDS No IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product (...) | TCCA STC No. SA05-60 Issue 1 dated 14 June 2005.Aero Consulting Services Ltd Configuration Definition List E025000Revision ER dated 24 March 2011.Bombardier Aerospace Aeroplane Flight Manual Supplement G-FM25004501 Revision N/C.GC94080000-IFCA Instru (...) | Restricted to ASN 9408. | 13.04.2011 |
| 10034902 | BOMBARDIER INC. | | | TCCA STC C-LSA11-024/D | BD-700-1A11 | Installation, Antenna Provisions | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product areuncha (...) | Transport Canada STC No. C-LSA11-024/D Issue 1 dated 02 May2011ACS-NAI Ltd. Modification Data Summary G731000 Rev NC or later approvedrevisionSupplemental Airworthiness Limitations, G731000-AWL Revision IR or laterapproved revisionInstructions for C (...) | None | 11.05.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------|--------------|----------------|----------------------------|----------------------------|--|---|---|---|---------------|
| 10035585 | BOMBARDIER INC. | | REV. 1 | TCCA STC NO. C-LSA02-261/D | BD-700-1A10 | Approval of Side Facing Divan Seating Limitations | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | Transport Canada STC No. C-LSA02-261/D Issue 2 dated 12 November 2003.Aero Consulting Services Modification Data Summary Number D04100Revision A or later approved revision.Aero Consulting Services Flight Manual Supplement D04191 Revision Aorlater ap (...) | Restricted to ASN 9054.Previous embodiment of EASA STC No. 10016140 (FAA STC No SA8164NM-D) | 06.07.2011 |
| 10035701 | BOMBARDIER INC. | | REV. 1 | TCCA STC NO. SA11-30 | BD-700-1A10 BD-700-1A11 | Installation, Portable Water System UV Filter | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | Transport Canada STC No. SA11-30 Issue 1 dated 13 April 2011.ACS-NAI Ltd. Modification Data Summary G560000 Revision NCor later approved revision.ACS-NAI Ltd. Instructions for Continued Airworthiness G560070-ICARRevision NC or later approved revision. (...) | Prior to the installation of this modification it must be determinedthat the aircraft interior has been approved by theissuance of an EASASTC that corresponds to one of the Transport Canada STCs listed onTransport Canada STC No. SA11-30. | 14.07.2011 |
| 10035704 | BOMBARDIER INC. | | | TCCA STC NO. C-LSA11-025/D | BD-700-1A11 | Installation, Avionics Systems | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | Transport Canada STC No. C-LSA11-025/D Issue 1 dated 07 July 2011.ACS-NAI Ltd. Modification Data Summary G732000 Revision NC or later approved revision.Bombardier Aerospace Flight Manual Supplement for V/UHF Radio, TACAN andDual Mode S/IFF Transponde (...) | 1. Restricted to ASNs 9395, 9404, 9411, 94172. Embodiment of EASA STC No. 10034902 (Transport Canada STC No.C-LSA11-024/D) | 13.07.2011 |
| 10035711 | BOMBARDIER INC. | | | TCCA STC NO. C-LSA11-022/D | BD-700-1A11 | Installation, Complete Custom Aircraft Interior - ASNs 9395, 9404, 9411,9417 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | Transport Canada STC No. C-LSA11-022/D Issue 1 dated 13 May2011.ACS-NAI Ltd. Modification Data Summary G729000 Revision NC or later approved revision.Bombardier Aerospace AirplaneFlight Manual Supplement for InteriorCompletion, G-FM25004503 Revision (...) | 1. Restricted to ASNs 9395, 9404, 9411, 9417.2. Embodiment of EASA STC No. EASA.IM.A.S.02172 | 13.07.2011 |
| 10035712 | BOMBARDIER INC. | | | TCCA STC NO. C-LSA11-023/D | BD-700-1A11 | Installation, Medical Evacuation Configuration Provisions | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | Transport Canada STC No. C-LSA11-023/D Issue 1 dated 13 May2011.ACS-NAI Ltd. Modification Data Summary G730000 Revision NC or later approved revision.Bombardier Aerospace Flight Manual Supplement for the Provisions forMedical Evacuation Configuration (...) | 1. Restricted to ASNs 9395, 9404, 9411, 9417.2. Embodiment of EASA STC No. 10035711 (refers to PN 0010007349)(TransportCanada STC No. C-LSA11-022/D) | 13.07.2011 |
| 10035777 | BOMBARDIER INC. | | | TCCA STC NO. SA05-60 | BD-700-1A11 | Installation, Complete Custom Aircraft Interior - ASN 9418 | EASA TCDS No IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product (...) | TCCA STC No. SA05-60 Issue 1 dated 14 June 2005.Aero Consulting Services Ltd Configuration Definition List E025000Revision EV dated 15 June 2011.Bombardier Aerospace Aeroplane Flight Manual Supplement G-FM25004501Revision N/C or later approved revis (...) | Restricted to ASN 9418 | 19.07.2011 |
| 10036054 | BOMBARDIER INC. | | REV. 1 | TCCA STC SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9406 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | TCCA STC SA04-113 Issue 1 dated 28 October 2004.Aero Consulting Services Configuration Definition List D954000 RevisionHL dated 13 July 2011Instructions for Continued AirworthinessG94060000-IFCAor later revisions of the above listed documents approv (...) | Restricted to ASN 94061. Refer to the Limitations sections of the required AFM Supplement andto the referenced FCOMS2. WLAN system is compatible only to IEEE 802.11(b) | 08.08.2011 |
| 10036058 | BOMBARDIER INC. | | | TCCA STC SA05-60 | BD-700-1A11 | Installation, Complete Custom Aircraft Interior - ASN 9421 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | TCCA STC No. SA05-60 Issue 1 dated 14 June 2005.Aero Consulting Services Ltd Configuration Definition List E025000Revision FC dated 22 July 2011. Bombardier Aerospace Aeroplane Flight Manual Supplement G-FM25004501Revision N/C or later approved revis (...) | Restricted to ASN 9421 | 09.08.2011 |
| 10036060 | BOMBARDIER INC. | | REV. 1 | TCCA STC SA04-113 | BD-700-1A10 | Installation of cabin systems and custom furnishings applicable toBD-700-1A10 serial number 9393 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.Aero Consulting Services Configuration Definition List D954000 RevisionHH dated 22 June 2011.Instructions for Continued Airworthiness G93930000-IFCAor later revisions of the above listed document (...) | Restricted to ASN 93931. Refer to the Limitations sections of the required AFM Supplement andto the referenced FCOMS2.WLAN system is compatible only to IEEE 802.11(b). | 09.08.2011 |
| 10036095 | BOMBARDIER INC. | | | TCCA STC C-LSA01-452/D | BD-700-1A10 | Interior Modification - ASN 9020 - Installation, Modifications to APUBay Light, Galley Oven, Smoke Detector Wiring and Equipment BayProvisions | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | Transport Canada STC No C-LSA01-452/D Issue 3 dated 02 June2011.Aero Consulting Services Modification Data Summary Number D21500Revision A dated 02 June 2011 or later approved revision.or later revisions of the above listed documents approved by EAS (...) | Restricted to ASN 9020.Previous embodiment of EASA STC No. 10016140 (refers to PN 0010011877)(Installation of a Complete Custom Aircraft Interior in accordance withFAA STC SA8164NM-D) is mandatory. | 10.08.2011 |
| 10036096 | BOMBARDIER INC. | | | TCCA STC Q-LSA02-065/D | BD-700-1A10 | Miscellaneous Cabin Modifications | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | Transport Canada STC No Q-LSA02-065/D Issue 2 dated 05 August 2003.Aeroccas Master Document List MDL02164002 - Miscellaneous CabinModifications on Bombardier Global Express BD-700-1A10, ASN 9020.Revision A dated 16 July 2003.or later revisions of the (...) | Restricted to ASN 9020.Previous embodiment of EASA STC No. 10016140 (refers to PN 0010011877)(Installation of a Complete Custom Aircraft Interior in accordance withFAA STC SA8164NM-D) is mandatory. | 10.08.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------|--------------|----------------|-------------------------------|------------------------|--|--|--|--|---------------|
| 10036344 | BOMBARDIER INC. | | | TCCA STC NO. SA11-69 | BD-700-1A10, -1A11 | Installation, Structural Provisions for Shower Drain Feedthrough | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA11-69 Issue 1 dated 08 August 2011.ACS-NAI Ltd. Modification Data Summary G733000 Revision N/C or later approved revision.Bombardier Instructions for Continued Airworthiness G733000-ICA Revision R or later revision.Bombardier (...) | None | 02.09.2011 |
| 10036603 | BOMBARDIER INC. | | | TCCA STC SA05-60 | BD-700-1A11 | Installation, Complete Custom Aircraft Interior - ASN 9427 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. SA05-60 Issue 1, dated 14 June 2005.Aero Consulting Services Ltd Configuration Definition List E025000Revision FE, dated 20 September 2011.Bombardier Aerospace Aeroplane Flight Manual Supplement G-FM25004501,Revision N/C or later approve (...) | Restricted to ASN 9427 | 22.09.2011 |
| 10037040 | BOMBARDIER INC. | | | TCCA STC NO. SA04-113 ISSUE 1 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9413 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. SA04-113 Issue 1, dated 28 October 2004.Aero Consulting Services Configuration Definition List D954000 Revision JF, dated 06 October 2011.Instructions for Continued Airworthiness GC94130000-IFCA or later revisions of the above listed doc (...) | Restricted to ASN 9413 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS.2. WLAN System is compatible only to IEEE 802.11(b). | 26.10.2011 |
| 10037055 | BOMBARDIER INC. | | | TCCA STC NO. SA04-113 ISSUE 1 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9238 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. SA04-113 Issue 1, dated 28 October 2004.Aero Consulting Services Configuration Definition List D954000 Revision JH, dated 07 October 2011.Instructions for Continued Airworthiness GC92380000-IFCA or later revisions of the above listed doc (...) | Restricted to ASN 9238 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS.2. WLAN System is compatible only to IEEE 802.11(b). | 27.10.2011 |
| 10037096 | BOMBARDIER INC. | | | TCCA STC NO. SA04-113 | BOMBARDIER BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9420. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. SA04-113 Issue 1, dated 28 October 2004.Aero Consulting Services Configuration Definition List D954000 Revision JG, dated 07 October 2011.Instructions for Continued Airworthiness GC94200000-IFCA or later revisions of the above listed doc (...) | Restricted to ASN 9420 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS.2. WLAN System is compatible only to IEEE 802.11(b). | 02.11.2011 |
| 10037122 | BOMBARDIER INC. | | | TCCA STC NO. C-LSA05-030/D | BD-700-1A11 | Aircraft Completion, 13 Passenger Interior Configuration. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. C-LSA05-030/D Issue 5 dated 01 November 2011.Aero Consulting Services Modification Data Summary D452001 revision D or later approved revision.Bombardier Aerospace Flight Manual Supplement G-FM25000121 Revision N/C or later appr (...) | Restricted to ASN 9130.Drawing B170044250045 Rev N/C 'W/D-MOD. TO WIRELESS CABIN NETWORK WIRINGPROVISION' disables the Wireless LAN system by disconnecting, capping/Stowing the WLU electrical connector. In lieu of conformity to drawing B17004420045 Rev (...) | 04.11.2011 |
| 10037223 | BOMBARDIER INC. | | | TCCA STC SA04-113 | BOMBARDIER BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9419. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. SA04-113 Issue 1, dated 28 October 2004.Aero Consulting Services Configuration Definition List D954000 Revision JU, dated 13 October 2011.Instructions for Continued Airworthiness GC94190000-IFCA or later revisions of the above listed doc (...) | Restricted to ASN 9419 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS.2. WLAN System is compatible only to IEEE 802.11(b). | 11.11.2011 |
| 10037297 | BOMBARDIER INC. | | | TCCA STC SA05-60 | BD-700-1A11 | Installation of Complete Custom Aircraft Interior - ASN 9430. | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. SA05-60 Issue 1, dated 14 June 2005.Aero Consulting Services Ltd Configuration Definition List E025000Revision FJ, dated 10 November 2011.Bombardier Aerospace Aeroplane Flight Manual Supplement G-FM25004501,Revision N/C or later approve (...) | Restricted to ASN 9430 | 17.11.2011 |
| 10037640 | BOMBARDIER INC. | | | TCCA STC NO. SA04-113 | BD-700-1A10 | Installation of Complete Custom Aircraft Interior - ASN 9423. | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. SA04-113 Issue 1, dated 28 October 2004.Aero Consulting Services Configuration Definition List D954000 Revision KB, dated 07 December 2011.Instructions for Continued Airworthiness GC94230000-IFCA or later revisions of the above listed doc (...) | Restricted to ASN 9423 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS.2. WLAN System is compatible only to IEEE 802.11(b). | 12.12.2011 |
| 10037644 | BOMBARDIER INC. | | | TCCA STC SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9429. | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. SA04-113 Issue 1, dated 28 October 2004.Aero Consulting Services Configuration Definition List D954000 Revision JU, dated 30 November 2011.Instructions for Continued Airworthiness GC94290000-IFCA or later revisions of the above listed doc (...) | Restricted to ASN 9429 and 1. Refer to the Limitations section of the required AFM Supplement and the referenced FCOMS.2. WLAN System is compatible only to IEEE 802.11(b). | 13.12.2011 |
| 10037675 | BOMBARDIER INC. | | | TCCA STC NO. SA05-60 ISSUE 1 | BD-700-1A11 | Installation, Complete Custom Aircraft Interior - ASN 9226 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA STC No. SA05-60 Issue 1, dated 14 June 2005.Aero Consulting Services Ltd Configuration Definition List E025000Revision FF, dated 23 September 2011.Bombardier Aerospace Aeroplane Flight Manual Supplement G-FM25004501,Revision N/C or later approve (...) | Restricted to ASN 9226 | 14.12.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------|--------------|----------------|--------------------------------|--------------------------|---|---|---|--|---------------|
| 10038659 | BOMBARDIER INC. | | REV. 2 | TCCA STC SA12-24 | BD-700-1A10, BD-700-1A11 | Installation of Class II Electronic Flight Bag (EFB) System for Global6000 and / or Global 5000 aircraft. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA12-24 Issue 3, dated 15 August 2012; Bombardier Inc. Modification Data Summary Number RAZ-BM700-003 Revision A, approved 02 March 2012; Bombardier Airplane Flight Manual Supplement G-FM46054001 Revision Basic; Instructions for C (...) | None. | 08.11.2012 |
| 10038764 | BOMBARDIER INC. | | | TCCA STC C-LSA12-009/D | BD-700-1A11 | Installation of Complete Custom Aircraft Interior - ASN 9401 | EASA.IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged (...) | Transport Canada STC No. C-LSA12-009/D Issue 1, dated 29 February 2012. Bombardier Inc. Modification Data Summary RAZ-BM700-9401-MDS Revision A or later approved revision. Bombardier Inc Airplane Flight Manual Supplement for the Aircraft Completion. (...) | Restricted to ASN 9401 | 12.03.2012 |
| 10038864 | BOMBARDIER INC. | | | TCCA STC SA11-96 | BD-700-1A10 | Installation of a Frequency Converter Fuse | EASA.IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged (...) | Transport Canada STC No. SA11-96 Issue 2, dated 17 January 2012. Altitude Aerospace Ltd. Modification Summary No. MS240064, Revision NC, dated 21 November 2011, or later approved revision. Altitude Aerospace Ltd. Instructions for Continued Airworthiness (...) | None | 20.03.2012 |
| 10039764 | BOMBARDIER INC. | | | TCCA STC SA12-16, ISSUE 2 | BD-700-1A10 | Installation, Aircraft Completion Modifications ASN 9440. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA12-16 Issue 2 dated 20 March 2012; Bombardier Configuration Definition Record RAZ-BM700-001 Revision Edated 09 May 2012; Bombardier Airplane Flight Manual Supplement G-FM25054001 Revision Basic; Instructions for Continued Airwo (...) | Restricted to ASN 9440. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis. | 22.05.2012 |
| 10039790 | BOMBARDIER INC. | | | TCCA STC Q-LSA00-009/D ISSUE 1 | BD-700-1A10 | Transport Canada STC No. Q-LSA00-009/D Installation of Interior and Avionics. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. Q-LSA00-009/D Issue 1 dated 23 June 2000. Modification List ML100569 for the Interior and Avionics Installation into Bombardier Global Express BD-700-1A10, Revision B, dated 22 June 2000., BD700-BCCM-546 Rev A - FMS - Gaseous (...) | Restricted to ASN 9028. Installation of EASA STC No. 10039791 is a pre-requisite. | 23.05.2012 |
| 10039791 | BOMBARDIER INC. | | | TCCA STC C-LSA12-040/D ISSUE 1 | BD-700-1A10 | Installation, Modifications to Interior Completion - ASN 9028. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. C-LSA12-040/D Issue 1 dated 10 February 2012. ACS-NAI Ltd. Modification Data Summary H224000 Revision NC. Instructions for Continued Airworthiness document 9028-ICA Revision NC or later revisions of the above listed documents (...) | Restricted to ASN 9028. Installation of EASA STC No. 10039790 is a pre-requisite. | 23.05.2012 |
| 10039792 | BOMBARDIER INC. | | | TCCA STC LSA01-329 ISSUE 2 | BD-700-1A10 | Approval of Seating Limitations for Center & Aft Positions of Air RHDivan (FS 663). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. C-LSA01-329 Issue 2 dated 01 August 2001. Aero Consulting Services Modification Data Summary Number D00600 Revision N/C. Aero Consulting Services Flight Manual Supplement D00691 Revision N/C or later revisions of the above list (...) | Restricted to ASN 9028. Previous installation of EASA STC No. 10039790 and EASA STC No. 10039791 are pre-requisites. The aircraft is limited to private use only, unless the conditions specified in Flight Manual Supplement D00691 Revision N/C are satisfied. | 23.05.2012 |
| 10039808 | BOMBARDIER INC. | | | TCCA STC SA12-16, ISSUE 2 | BD-700-1A11 | Installation, Aircraft Completion Modifications ASN 9453 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA12-16 Issue 2 dated 20 March 2012; Bombardier Configuration Definition Record RAZ-BM700-001 Revision Edated 04 May 2012; Bombardier Airplane Flight Manual Supplement G-FM25054001 Revision Basic; Instructions for Continued Airwo (...) | Restricted to ASN 9453. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis. | 24.05.2012 |
| 10040171 | BOMBARDIER INC. | | | TCCA STC NO. SA12-16 | BD-700-1A10 | Installation, Aircraft Completion Modifications ASN 9450 | EASA.IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged (...) | Transport Canada STC No. SA12-16 Issue 2 dated 20 March 2012; Bombardier Configuration Definition Record RAZ-BM700-001 Revision Edated 05 June 2012; Bombardier Airplane Flight Manual Supplement G-FM25054001 Revision Basic; Instructions for Continued Airwo (...) | Restricted to ASN 9450. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis. | 15.06.2012 |
| 10040172 | BOMBARDIER INC. | | | TCCA STC NO. SA12-16 | BD-700-1A10 | Installation, Aircraft Completion Modification - ASN 9441 | EASA.IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged (...) | Transport Canada STC No. SA12-16 Issue 2 dated 20 March 2012; Bombardier Configuration Definition Record RAZ-BM700-001 Revision Edated 31 May 2012; Bombardier Airplane Flight Manual Supplement G-FM25054001 Revision Basic; Instructions for Continued Airwo (...) | Restricted to ASN 9441. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis. | 15.06.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------|-------------------|----------------|--------------------------|-------------------|---|--|--|--|---------------|
| 10040256 | BOMBARDIER INC. | | | TCCA STC NO. SA12-16 | BD-700-1A10 | Installation, Aircraft Completion Modifications ASN 9455. | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA12-16 Issue 2 dated 20 March 2012. Bombardier Configuration Definition Record RAZ-BM700-001 Revision N dated 12 June 2012. Bombardier Airplane Flight Manual Supplement G-FM25054001 Revision BASIC. Instructions for Continued Air (...) | Restricted to ASN 9455. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis. | 21.06.2012 |
| 10040296 | BOMBARDIER INC. | | | TCCA STC NO. SA12-16 | BD-700-1A10 | Installation of Aircraft Completion Modifications - ASN 9437. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA12-16 Issue 2 dated 20 March 2012. Bombardier Configuration Definition Record RAZ-BM700-001 Revision L dated 11 June 2012. Bombardier Airplane Flight Manual Supplement G-FM25054001 Revision BASIC. Instructions for Continued Air (...) | Restricted to ASN 9437. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis. | 22.06.2012 |
| 10040478 | BOMBARDIER INC. | | | TCCA STC NO. SA12-16 | BD-700-1A11 | Installation, Aircraft Completion Modifications ASN 9457 | EASA.IM.A.009 | Transport Canada STC No. SA12-16 Issue 2, dated 20 March 2012. Bombardier Configuration Definition Record RAZ-BM700-001 Revision H, dated 28 May 2012. Bombardier Airplane Flight Manual Supplement G-FM25054001 Revision BASIC. Instructions for Continued Air (...) | Restricted to ASN 9457. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis. | 05.07.2012 |
| 10040601 | BOMBARDIER INC. | | | TCCA STC SA12-16 ISSUE 2 | BD-700-1A11 | Installation, Aircraft Completion Modifications ASN 9461 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA12-16 Issue 2, dated 20 March 2012. Bombardier Configuration Definition Record RAZ-BM700-001 Revision M, dated 11 June 2012. Bombardier Airplane Flight Manual Supplement G-FM25054001 Revision BASIC. Instructions for Continued Air (...) | Restricted to ASN 9461. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin portable electronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis. | 12.07.2012 |
| 10040720 | BOMBARDIER INC. | | | TCCA STC NO. SA12-16 | BD-700-1A10 | Installation, Aircraft Completion Modifications ASN 9443. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA12-16 Issue 2 dated 20 March 2012. Bombardier Configuration Definition Record RAZ-BM700-001 Revision T dated 10 July 2012. Bombardier Airplane Flight Manual Supplement G-FM25054001 Revision BASIC. Instructions for Continued Air (...) | Restricted to ASN 9443. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis. | 24.07.2012 |
| 10040722 | BOMBARDIER INC. | | | TCCA STC NO. SA12-31 | BD-700-1A10/-1A11 | Modification, Electronic Wiring Heater Current / Brake Temperature Monitor Unit (HBMU). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA12-31 Issue 2 dated 25 May 2012. ACS-NAI Ltd. Modification Data Summary H119000 Revision N/C, dated 30 March 2012; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision). | None. | 24.07.2012 |
| 10041031 | BOMBARDIER INC. | | | TCCA STC NO. O-LSA02-400 | BD-700-1A10 | Chairman Interior Installation | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. O-LSA02-400 Issue 6 dated 29 June 2012. DECA Aviation Engineering Limited Modification Summary MS01050, Revision 7 dated 28 April 2003. Aero Consulting Services Ltd. Modification Data Summary G151000 Revision N/C. ACS-NAI Ltd. Mo (...) | Restricted to ASN 9003. Previous embodiment of JAA (EASA) Accepted Bombardier Service Bulletin 700-23-001, Satellite Communications System - Installation is prerequisite. The Public Address System does not comply with requirements of JAR25.1423. It is no (...) | 16.08.2012 |
| 10041238 | BOMBARDIER INC. | EASA.IM.A.S.01208 | REV. 1 | TCCA STC C-LSA05-104/D | BD700-1A10 | Cabin Interior Completion / Furnishings | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Ref.: Limited Supplemental Type Certificate by TCCA - No. C-LSA05-104/DFM-S: G-FM25000127 A/A/LI; G-TM05100066 N/C/FCOM-S: G-MS25000019 N/C/ICA; GC91470001-IFCA or later revisions of the above listed documents approved by EASA in accordance with EASA (...) | Restricted to ASN 9147 | 17.09.2012 |
| 10041239 | BOMBARDIER INC. | EASA.IM.A.S.01230 | REV. 1 | TCCA STC C-LSA05-169/D | BD700-1A10 | Passenger Occupancy of the Crew Rest Area | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Ref.: Limited Supplemental Type Certificate by TCCA - No. C-LSA05-169/DFCOM-S: G-MS25000020 N/C or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision). | Restricted to ASN 9147 | 17.09.2012 |
| 10041267 | BOMBARDIER INC. | | REV. 1 | TCCA STC SA03-113 | CL-600-2B16 | Installation of Complete Customized Aircraft Interior. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Aero Consulting Services Ltd. Configuration Definition List ref. D817000 dated 19 Jan 2007, Revision AD; Aero Consulting Services Ltd. Airplane Flight Manual Supplement FMSD817000 rev A; Mandatory Maintenance Manual Supplement ref. D817070. Rev (...) | This STC is limited to aircraft Bombardier CL-600-2B16 (604variant), Serial Number 5580, 5589, 5591, 5628, 5636 and 5655. | 14.12.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|---------------------|----------------|---------------------------|----------------------------|---|--|--|--|---------------|
| 10041344 | BOMBARDIER INC. | | | TCCA STC SA12-16 | BD-700-1A10 | Installation, Aircraft Completion Modifications ASN 9456. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA12-16 Issue 2 dated 20 March 2012. Bombardier Configuration Definition Record RAZ-BM700-001 Revision V dated 27 August 2012. Bombardier Airplane Flight Manual Supplement G-FM25054001 Revision BASIC. Instructions for Continued A (...) | Restricted to ASN 9456. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis. | 10.09.2012 |
| 10041500 | BOMBARDIER INC. | | REV. 1 | TCCA STC SA-12-16 ISSUE 2 | BD-700-1A10 | Installation, Aircraft Completion Modifications ASN 9464 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA-12-16 Issue 2, dated 20 March 2012. Bombardier Configuration Definition Record RAZ-BM700-001 Revision AD, dated 14 September 2012. Bombardier Airplane Flight Manual Supplement G-FM25054001 Revision BASIC. Instructions for Conti (...) | Restricted to ASN 9464 | 24.09.2012 |
| 10041512 | BOMBARDIER INC. | | | TCCA STC SA-12-16 ISSUE 2 | BD-700-1A11 | Installation, Aircraft Completion Modifications ASN 9462 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA-12-16 Issue 2, dated 20 March 2012. Bombardier Configuration Definition Record RAZ-BM700-001 Revision AB, dated 06 September 2012. Bombardier Airplane Flight Manual Supplement G-FM25054001 Revision BASIC. Instructions for Contin (...) | Restricted to ASN 9462. | 24.09.2012 |
| 10041975 | BOMBARDIER INC. | | | TCCA STC NO. SA12-16 | BD-700-1A11 | Installation, Aircraft Completion Modifications ASN 9458. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA12-16 Issue 2, dated 20 March 2012. Bombardier Configuration Definition Record RAZ-BM700-001 Revision U, dated 24 July 2012. Bombardier Airplane Flight Manual Supplement G-FM25054001 Revision BASIC. Instructions for Continued A (...) | Restricted to ASN 9458. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis. | 29.10.2012 |
| 10042314 | BOMBARDIER INC. | | REV. 1 | TCCA STC NO. SA12-16 | BD-700-1A10 BD-700-1A11 | Installation, Aircraft Completion Modifications EASA, I.M.A.009 Bombardier BD-700-1A10 & BD-700-1A11 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS25 Amendment 1 CRI D-100 Issue 2, Doors Between Passenger Compartments CRI D-200 Issue 2, Side Facing Divans CRI D-300 Iss (...) | Transport Canada STC No. SA12-16 Issue 2, dated 20 March 2012. Bombardier Configuration Definition Record RAZ-BM700-001 Revision BM, dated 8th February 2013. Bombardier Airplane Flight Manual Supplement G-FM25054001 Revision BASIC. Instructions for Contin (...) | EASA approved configurations are identified in Configuration Definition Record RAZ-BM700-001. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic device (...) | 23.08.2013 |
| 10042835 | BOMBARDIER INC. | | | STC NO C-LSA03-348/D | BD-700-1A10 | Aircraft Completion, 13 Passenger Interior Configuration | EASA, I.M.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged (...) | Transport Canada STC No. C-LSA03-348/D Issue 2 dated 19 January 2004. Aero Consulting Services Modification Data Summary D49100 Revision A. Bombardier Aerospace Flight Manual Supplements: G-FM24000044 Rev N/C Cabin Master Power Switch G-FM25000072 Rev A (...) | Restricted to ASN 9093. Bombardier Service Bulletin 700-31-018 Modification - Integrated Avionics Computers - Avionics 2001 Batch 1 IAC Upgrade must be embodied. | 20.12.2012 |
| 10043057 | BOMBARDIER INC. | EASA, I.M.A.S.03216 | | TCCA STC SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9283 | | Transport Canada SoC reference C-09-0274 dated 12 June 2009. TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D954000 Revision DA dated 29 May 2009 or later approved revision. Bombardier Aerospace (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The i (...) | 16.06.2009 |
| 10043058 | BOMBARDIER INC. | EASA, I.M.A.S.03163 | | TCCA STC SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9294 | | - Transport Canada SoC reference C-09-0213 dated 15 May 2009. - TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. - Aero Consulting Services Configuration Definition List D954000 Revision CL, dated 20 March 2009 - Instructions for Continued Airworthiness GC92940000-IFCA. | 1. Restricted to ASN 9294.2. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS.3. WLAN system is compatible only to IEEE 802.11(b), 4. Prior to installation of this modification the installer must determine th (...) | 02.06.2009 |
| 10043375 | BOMBARDIER INC. | | | TCCA STC NO. C-LSA12-011 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9381 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI D-600 Shower Installation The requirements for environmental protection and the associated certified noise and/or emis (...) | TCCA STC No. C-LSA12-011/D Issue 1 dated 17 February 2012. Bombardier Inc. Modification Data Summary RAZ-BM700-9381-MDS Revision NC Airplane Flight Manual Supplement for the Aircraft Completion G-FM25060001 Revision NC. Instructions for Continued Airwo (...) | Restricted to ASN 9381. Installation of structural provisions for the shower drain feed through in accordance with EASA Supplemental Type Certificate number 10036344 (TCCA STC No. SA11-69) is a prerequisite for this STC. Prior to installation of this des (...) | 24.01.2013 |
| 10044180 | BOMBARDIER INC. | | REV. 1 | TCCA STC SA04-113 | BD-700-1A10 | Installation, Complete Custom Aircraft Interior | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): CRI D-200 Issue 2, Side Facing Divans Equivalent Safety Finding(s): CRI D-500 Issue 2, Emergency Exi (...) | Transport Canada STC No. SA04-113 Issue 1, dated 28 October 2004. ACS-NAI Configuration Definition List D954000 Revision LP, dated 07 March 2013. Airplane Flight Manual Supplement G-FM25004001 Revision NC-Mandatory Maintenance Manual Supplement G-TM05 (...) | The EASA Approved Configurations are identified in ACS-NAI Configuration Definition List D954000. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic dev (...) | 15.04.2013 |
| 10016316 | BOMBARDIER INC. - COMPLETION CENTRE | EASA, I.M.A.S.02622 | REV. 7 | TCCA STC SA07-119 | BD100 CHALLENGER 300 | Installation, In-Service Improvement Modifications In-service design improvements to cabin interior features Revision of the EASA STC 10016316, Rev. 6 due to revised technical data (Modification Data Summary F344000 Rev. H) | EASA TCDS I.M.A.080 Issue 3, 26th October 2010 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels (...) | Transport Canada Supplemental Type Certificate SA07-119 Issue 2, dated 20th October 2010. - Transport Canada Approved Aero Consulting Services Ltd. Modification Data Summary F344000 Revision H, dated 28th September 2010 or later revisions of the above (...) | Not applicable | 28.10.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|-------------------|----------------|----------------------|---|--|---|---|---|---------------|
| 10029547 | BOMBARDIER SHORTS BROTHERS, PLC | P-EASA.A.S.03579 | | | A319, A320, A321 SERIES | V2500 NACELLE - RTI MONOLITHIC FAN COWL DOOR | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ..., Certification Compliance Checklist, ref. doc. SB-AWO-ESTC-001, Revision NC, 11th December 20092., Master Data List, ref. doc. SB-V335-037, Revision H, 10th March 20103., Maintenance Manual and Instructions for Continued Airworthiness, ref. doc. 71 (...) | None. | 31.03.2010 |
| 10030474 | BOND HELICOPTERS EUROPE Ltd. | | | | EC135 T2(CPDS), EC135 T2+ | Engine Usage Monitoring System Installation of an Engine Usage Monitoring System in accordance with change number BOND/010 as defined in DL/010-099 to monitor certain engine parameters, further described in Design Report DR/010-001. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CS 27 Amdt. 2 for the Change The requirements for environmental protection and the associa (...) | Flight Manual Supplement EC135 T2, T2+ "Engine Usage Monitoring System", FMS/010-002, Issue 1 for later revisions of the above listed documents approved by EASA - Post installation Actions PIA/010-009 Iss. 1- Instructions for Continued Airworthiness IC (...) | None | 17.06.2010 |
| 10031949 | BOND HELICOPTERS EUROPE Ltd. | | REV. 1 | | AS 332 L2 | Installation of TCAS II system in accordance with BOND/010. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CS 29 Amendment 2 Special Condition F-01: Protection from the effects of HIRF, dated 21 (...) | Flight Manual Supplement FMS/018-002 Issue 1 for later revisions of the above listed documents approved by EASA - Design Report DR/018-001 Issue 6- Post Installation Actions PIA/018-011 Issue 2- Instructions for continued airworthiness ICA/018-012 Issue 2- Drawing List DL/018-099 Issue 7 | None | 06.07.2011 |
| 10039850 | BOND HELICOPTERS EUROPE Ltd. | | | | EC 135 P1 EC 135 P2 EC 135 P2+ EC 135 T1 EC 135 T2 EC 135 T2+ EC 635 P2+ EC 635 T1 EC 635 T2+ | BOND/060 "Martin- Baker Rotating Seat Installation, EC 135" | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CS 27 Amendment 2 The requirements for environmental protection and the associated certi (...) | Rotorcraft Flight Manual Supplement FMS/060-002, dated May 3rd, 2012; or later revisions of the above listed documents approved by EASA - Design Report DR/060-001, Issue 1, dated May 3rd 2012 - Drawing List DL/060-099, Issue 1- Instructions for Continued Airworthiness ICA/060-008, Issue 1 | None. | 29.05.2012 |
| 10042728 | BOND HELICOPTERS EUROPE Ltd. | | | | EC135 T1,-T2,-T2+,-P1,-P2,-P2+ | "East Anglia Air Ambulance Role Equipment, EC 135" for installation of NVIS equipment based on modification BOND/063 as defined in DL/063-099. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CS 27 Amdt. 2 for 27.1, .21, .251, .301, .303, .305, .321, .337, .341, .351, .561, .601 (...) | Rotorcraft Flight Manual Supplement "NIGHT VISION IMAGING SYSTEM, POWERLINE DETECTION SYSTEM, ENGINE USAGE MONITORING", FMS/063-002 Issue 1 for later revisions of the above listed documents approved by EASA - Design Report DR/063-001, Issue 6- Drawing L (...) | Limited to Aircraft with CPDS and FCDS installed - This STC applies only to the models listed above. This rotorcraft employs Night Vision Imaging System lighting, commonly named NVISighting, that has been certified to ensure the aircraft is compatible (...) | 14.12.2012 |
| 10044696 | BOND HELICOPTERS EUROPE Ltd. | | | | EC 635 T1 (CPDS), EC135 P1 (CPDS), EC135 P2+ EC135 P2 (CPDS), EC135 T2+ EC135 T1 (CPDS), EC 635 P2+ EC135 T2 (CPDS), EC 635 T2+ | Trakka Searchlight on NVIS EC 135 | CS 27 Amendment 3 The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement FMS/076-002, Issue 1 for later revisions of the above listed documents approved by EASA - Design Report DR/076-001, Issue 4- Modification Leaflet & Certificate of Design BOND/076, Issue 4- Instructions for Continued Airworthiness ICA/076-012, Issue 2 | None | 30.04.2013 |
| 10017028 | BOUNDARY LAYER RESEARCH INC | EASA.IM.R.S.01447 | | FAA STC SR01208SE | BELL 206/A/B/L/L-1/L-3/L-4 | Installation of dual tailboom strakes. This STC has been classified as #BASIC# by the FAA and has been approved under the PTVP. | | See attached sheets | | 04.04.2008 |
| 10014082 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.01174 | | | BOEING 737-200 | Revised Seating Layout of 124 Seats and 130 Seats using BAC1288 Series Seats | | -Bournemouth Aviation (Consultants) Compliance Document BAC/737/035 dated March 3 2005 or later approved revision | | 17.03.2005 |
| 10014083 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.01175 | | | BOEING 737-200 | Deactivation of Type III Exits & Installation of a 60 Seat VIP Passenger configuration. | | - Bournemouth Aviation Consultants Compliance Document Ref.: BAC/737/036 Top Drawing BAC 1289-01 | | 03.03.2005 |
| 10014277 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.01513 | REV 2 | | BAE 146 BAE 146/AVRO 146 SERIES BAE 146/RJ SERIES | Installation of PRNAV Compliant UNS-1E FMS | | - Bournemouth Aviation (Consultants) Modification BAC/146/062 Issue 1 or later approved revision, - Bournemouth Aviation (Consultants) Modification BAC/146/068, - Bournemouth Aviation (Consultants) AFM Supplement FMS/345/30 Issue 2 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.12.2006 |
| 10014310 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.01555 | | | BAE 146 | Change EGPWS Altitude Callouts | | -Bournemouth Aviation (Consultants) Ltd Modification BAC/146/057 Issue 1 or later approved revision. | | 16.03.2006 |
| 10014331 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.01585 | | | BOEING 737-200 | Installation of Cargo Hold Smoke Detection & Fire Suppression System | | Bournemouth Aviation (Consultants) Compliance Document ref.BAC/737/046 issue 1 or alter approved revision | | 01.12.2005 |
| 10014352 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.01617 | | | BAE 146/AVRO 146 SERIES | Installation of BRNAV Compliant GPS | | Bournemouth Aviation (Consultants) Modification BAC/146/064 Issue 1 or later approved revision Bournemouth Aviation (Consultants) AFM supplement FMS/353/31 Issue 1 dated 25 November 2005 or later approved revision | | 12.12.2005 |
| 10014437 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02487 | | | BAE 146-200 | Layout of Passenger Accommodation (LOPA) for 84, 93, 95 & 96 | | Bournemouth Aviation (Consultants) Ltd Modification BAC/146/066 | | 19.04.2006 |
| 10014438 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02488 | | | BOEING 737-400 | Installation of 136 (161/120) PAX LOPA | | Bournemouth Aviation (Consultants) Ltd Modification BAC/373/049 | | 07.04.2006 |
| 10014467 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02528 | | | B757-200 | Layout of Passenger Accommodation, C2C Seats with new audioIFE & Seat Covers. | | Bournemouth Aviation (Consultants) BAC/757/005 | | 20.03.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|----------------|----------------|----------------------|--|--|---|---|---|---------------|
| 10014473 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02534 | | | BOEING 757-200 | Layout of Passenger Accommodation - 92 Business & 8 Economy/Crew | | Bournemouth Aviation (Consultants) BAC/757/006 | | 20.03.2006 |
| 10014478 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02540 | | | CESSNA 441 SERIES | Installation of Two L Band Antennae | | Bournemouth Aviation (Consultants) Definition Document ref J379/DD/01 Issue 1 dated March 08 2006 | Installation of Aircraft Registered within EASA Member State only | 17.05.2006 |
| 10014532 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02618 | | | BOMBARDIER DHC-8-100 SERIES | ACS Change 7 Upgrade & Elementary Surveillance Upgrade | | Bournemouth Aviation (Consultants) Ltd BAC/DHC8/006Bournemouth Aviation (Consultants) Ltd AFM Supplement FMS/397/35 Issue 1 dated 15 June 2006 or later approved revisions | -- | 04.08.2006 |
| 10014533 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02620 | | | BOMBARDIER DHC-8-100 SERIES | Installation of HF System | | Bournemouth Aviation (Consultants) Ltd BAC/DHC8/004 | -- | 04.08.2006 |
| 10014534 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02621 | REV 1 | | BOMBARDIER DHC-8-102 BOMBARDIER DHC-8-103 | Installation of Dual UNS-1E Flight Management System.Note: Rev.1 to STC EASA.A.S.02621 adds RNP Approach Capability. | | Bournemouth Aviation (Consultants) Ltd Doc. BAC/DHC8/001, Iss. 6 or later approved revision. - Bournemouth Aviation (Consultants) Ltd Flight Manual Supplement FMS/397/33, Iss. 4 or later approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer.2. Prior (...) | 26.01.2009 |
| 10014535 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02622 | | | BOMBARDIER DHC-8-100 SERIES | Installation of Aircell ST3100 Satellite Telephone System (SATCOM) | | Bournemouth Aviation (Consultants) Ltd BAC/DHC8/003Bournemouth Aviation (Consultants) Ltd AFM Supplement FMS/397/36 Issue 1 dated July 6, 2006 or later approved revision | -- | 04.08.2006 |
| 10014536 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02623 | | | BOMBARDIER DHC-8-100 SERIES | Installation of Universal Terrain Awareness Warning System (TAWs) | | Bournemouth Aviation (Consultants) Ltd BAC/DHC8/003Bournemouth Aviation (Consultants) Ltd AFM Supplement FMS/397/34 Issue 1 dated June 15, 2006 or later approved revision | -- | 04.08.2006 |
| 10014588 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02695 | | | BOEING 737-200 | Installation of 106 Seat LOPA | | Bournemouth Aviation (Consultants) Modification BAC/737/056 | | 31.07.2006 |
| 10014596 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02713 | | | BOMBARDIER CL-600-2B16 (604) | Omnibus Modification to Support Aircraft Importation. | | Bournemouth Aviation (Consultants) Omnibus Modification BAC/CL/001. | Restricted to ASN 5383. | 28.11.2006 |
| 10014636 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02782 | | | AIRBUS A330 | Installation of Quick Access Recorder. | | Compliance Document Summary Sheet and Accomplishment Instructions Continuation Sheet, Ref BAC/ABUS/009, Issue 2 dated 19 December 2006. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description & Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 02.01.2007 |
| 10014640 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02789 | | | AEROSPATIALE ATR 42 AEROSPATIALE ATR 72 | Inraero Siren Cargo Container System installation on ATR42-320 and ATR72-202 aircraft previously converted to Class E Cargo Compartment Freighter. | | For installation on ATR42-320 aircraft:Bournemouth AviationBAC/ATR/003 Issue 2 Siren Cargo Loading System InstallationATR42 and# Installation Instructions as per Bournemouth Aviation Accomplishment Instructions BAC/ATR/003 Issue 2 andDrawing BAC-1522-01 Issue B.# Maintenance Instructions as | This STC is applicable on the following aircraft:Those ATR42-320 previously converted to Class E Cargo Compartment Freighter by the embodiment of the following ATR GIE Service Bulletins:# ATR42-25-0070 Reinforce all floor panels, and# ATR42-25-0063 Modify A/C for Cargo Container Loading, and# | 29.01.2008 |
| 10014697 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.02864 | | | BOEING 757-200 | Installation of B/C LOPA with standing bar tables | | Change Number J438CMM J438/PR/016 | | 08.03.2007 |
| 10014863 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.03087 | REV. 4 | | B747-100, -200B/C/F, -300, B747-400, -400F | STC EASA.A.S.03087: This modification introduces a Universal CockpitDisplay System (Electronic Flight Bag) into the aircraft consisting of 2Flight Deck Displays, an Application Server Unit, an Accessible DataUnit, an external video camera and assoc (...) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS-25 Amendment 2 for the areas of changeThe requirements for environmental protection (...) | FMS-468-38 Rev. 5 Airplane Flight Manual SupplementUniversal Avionics Systems Corporation Service Bulletin 340.06.02-3284Installation of SCN 32.2 and Mod 2 in the ASUor later revisions of the above listed documents approved by EASA. | None. | 21.05.2012 |
| 10014928 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.03181 | | | BAE 146-200 | SSR Mode S Enhanced Surveillance | | Mod No BAC/146/100 Issue 1AFM FMS/486/42 | | 16.08.2007 |
| 10014999 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.03299 | | | AIRBUS 1-11 | TAWs Installation. | | Flight Manual Supplement FMS/524/44 rev1Modification Document BAC/111/008 issues 4 and 5 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 30.09.2008 |
| 10015001 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.03301 | | | BAE 146 | Installation of VIP LOPA (Inc. Club 4 Layout). | | Modification BAC/146/108Instructions for Continued Airworthiness ICA/033/525 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 14.07.2008 |
| 10015020 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.03326 | | | BOEING 737-500 | Layout of Passenger Accommodation. | | BAC Modification BAC/737/082 | | 19.05.2008 |
| 10015023 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.03329 | | | BOEING 767-300 | Layout Passenger Accomodation | | Modification BAC/767/009. | | 17.04.2008 |
| 10015048 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.03364 | | | BOEING 757 - 200 | Installation of revised Passenger Accommodation | | Bournemouth Aviation (Consultants) Ltd Compliance Document Summary Sheet and Accomplishment Instructions, BAC/757/014, Issue 3, 22/01/2009.Instruction for Continued Airworthiness Boeing 757 188/200 PAX LOPA, ICA/046/533, Issue 1, 22/10/2008 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.The (...) | 23.01.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|-----------------|----------------|----------------------|------------------------------|---|--|---|---|---------------|
| 10015117 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.03466 | | | BOMBARDIER CL-600-2B16 (604) | Installation of Electronic Flight Bag | | BAC Modification No. BAC/CL023 Issue 2BAC FMS/535/45 Issue2 | Restricted to ASN 5383 | 06.06.2008 |
| 10015122 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.03471 | REV. 2 | | 737-300 | B1/B2 Engine Thrust Programming Connector Installation | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Compliance document BAC/737/091 Issue 5 dated 22 January 2013. Instructions for Continued Airworthiness For Group 1 aircraft refer to ICA/050/544 For Group 2 aircraft refer to BAC-544ICA01 or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 24.01.2013 |
| 10015290 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.A.S.03825 | REV. 1 | | A320 | Installation of 148 PAX Layout of Passenger Accommodation (LOPA) This STC is revised in order to remove the EWIS Limitation | - Type Certification Basis in accordance with the TCDS No. A.064 issue 8 that includes CRJ H-01 / Special Condition "Enhanced Airworthiness Programme for Aeroplane System -ICA on EWIS" The requirements for environmental protection and the associated dc (...) | - Compliance Document Summary Sheet, ref. doc. BAC/ABUS/020, Issue 5, dated 30th August 2012- Compliance Check List, ref. doc. CCL/001/628, Issue 5, dated 30th August 2012- Instruction for Continued Airworthiness, ref. doc. ICA/072/628, Issue 3, dated (...) | none | 12.11.2012 |
| 10017101 | BOURNEMOUTH AVIATION (CONSULTANTS) | EASA.R.S.01071 | | | MD900 | Upgrading of existing Single Cargo Hook to Dual Hook System | | -Bournemouth Aviation Consultants Modification BAC/MD/003-Bournemouth Aviation Consultants AFM Supplement FMS/253/28 Revision 1 or later approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 07.09.2005 |
| 10029836 | BOURNEMOUTH AVIATION (CONSULTANTS) | | | | BOEING 737-800 | 164 Passenger Layout. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Compliance Document BAC/737/131 Issue 7- Instructions for Continued Airworthiness ICA/114/681 Issue 1 or later revisions of the above listed documents approved by EASA | None. | 23.04.2010 |
| 10038818 | BOURNEMOUTH AVIATION (CONSULTANTS) | | REV. 1 | | AS332 L2 | Installation of Seat Rail Mounted Fuel Tank. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | BAC-761FMS-01 revision 1 dated 19 September 2011 - Flight Manual Supplement or later revisions of the above listed documents approved by EASA. BAC-761ACI01 issue 2 dated 19 September 2011 - Accomplishment Instructions: BAC-761CCL01 issue 4 dated 9 March 2 (...) | None. | 20.03.2012 |
| 10038904 | BOURNEMOUTH AVIATION (CONSULTANTS) | 0010012414-001 | | CAA UK STC AAN 27329 | A109E | OBSERVER MOVING MAP SYSTEM INSTALLATION The system is manufactured by Skyforce and consists of a 5" Display/Control Head, a Processor Unit, and a Control Panel. The equipment is CAA approved under miscellaneous approvals No. MIS00183. The installation is (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Sloane Flight Manual Supplement No.6 issue 2 or later revisions of the above listed documents approved by EASA. Modification Definition and Maintenance requirements: "Sloane Modification document P017/109/MO" | The approval holder shall fulfill the obligations of Part 21, Paragraph 21A.118A. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ (...) | 22.03.2012 |
| 10038905 | BOURNEMOUTH AVIATION (CONSULTANTS) | 0010012415-001 | | CAA UK STC AAN 27645 | A109E | OXYGEN SYSTEM MODIFICATION Replace (before) P/N 1270152-3 Oxygen cylinder with (now) "British Oxygen Company" (BOC), Medical Gases AF size cylinder. The installation is identified and defined by the "Sloane Helicopters Ltd. Modification document No. P023/109/MO Issue 2". | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Agusta Westland supplied RFM Appendix 10 for the Emergency Medical Service (EMS) remains applicable or later revisions of the above listed documents approved by EASA. Modification Definition and Maintenance requirements: "Sloane Modification document P023/109/MO" | None | 22.03.2012 |
| 10038906 | BOURNEMOUTH AVIATION (CONSULTANTS) | 0010012416-001 | | CAA UK STC AAN 27764 | A109E | Installation of Bendix/King RDR 2000 Weather Radar System (Bendix/King RDR 200 is CAA approved Ref. VC 01048). The installation consists of an antenna-receiver-transmitter (ART) unit, the Configuration Module, and the Control Panel. The (ART) radar un (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Sloane Flight Manual Supplement No.7 or later revisions of the above listed documents approved by EASA. Modification Definition and Maintenance requirements: "Sloane Modification document P036/109/MO Issue A" | None | 22.03.2012 |
| 10038907 | BOURNEMOUTH AVIATION (CONSULTANTS) | 0010012417-001 | | CAA UK STC AAN 27763 | A109E | Trimble 2101 (IO) Approach Plus GPS (CAA approved Ref. VC01048) The installation consists of a Jeppesen data base slot card, a Receiver/Display Unit (RDU) interfaced to the EFIS via ARINC 429 and fitted on the top centre of the instrument panel. A GPS a (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Sloane Flight Manual Supplement No.8 or later revisions of the above listed documents approved by EASA. Modification Definition and Maintenance requirements: "Sloane Modification document P037/109/MO Issue A" | A placard is placed in clear view of the pilot and reads: GPS SYSTEM. USE ONLY DURING VFR OPERATIONS as specified in the Sloane Flight Manual Supplement No 8:1. The GPS systems to be used in VFR conditions only, for use as supplemental navigation system (...) | 22.03.2012 |
| 10038986 | BOURNEMOUTH AVIATION (CONSULTANTS) | | | | BOEING 757-200 | Installation of 21 Horse Stalls and a 10 Passenger Pallet. | The Certification Basis for the original product remains applicable to this certificate/ approval. CRJ H-01 EWIS The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged (...) | BAC Ltd Master Data List No. BAC-820MDL05 Iss 1 dated 27 March 2012 or later approved revision; or later revisions of the above listed documents approved by EASA. | This EASA STC does not approve the installation of Horse Boxes. The following BAC Minor Modifications must be installed as prerequisite for this approval:- Mod BAC-820MOD01- Mod BAC-820MOD02- Mod BAC-820MOD03- Mod BAC-820MOD04- Mod BAC-820MOD06 | 28.03.2012 |
| 10041170 | BOURNEMOUTH AVIATION (CONSULTANTS) | CAA-UK AAN28344 | REV. 1 | | A109A II | AVIONICS UPGRADE FOR A109A II Installation of 2 x Garmin GNS430 Nav/Com/GPS, 1 x Garmin GTX327 Transponder, 1 x Garmin GMA340 Audio Panel, SkyMap KMD150 and Kannad 406AF-H ELT. Also King KNI 582-01. Removal of 2 x Collins Comms, 2 x Collins Navs, Collin (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | SLOANE FLIGHT MANUAL SUPPLEMENT No. 11 or later revisions of the above listed documents approved by EASA. DOCUMENTATION OF MODIFICATION P055/109/MO | ONE PLACARD WITH THE WORDS "GPS LIMITED TO VFR ONLY" POSITIONED ON THE INSTRUMENT PANEL NEAR THE KMD 150 | 03.09.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|--------------------|----------------|--------------------------------|--|---|---|---|---|---------------|
| 10041983 | BOURNEMOUTH AVIATION (CONSULTANTS) | | | | 767-300 | Installation y 276 PAX LOPA on B767-300. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | BAC Ltd Master Data List No. BAC-784MDL02 Issue 4 dated 24 October 2012; or later revisions of the above listed documents approved by EASA. | The following BAC minor modifications are prerequisites for this modification: (1) Mod BAC-784MOD12 Revision of Emergency safety Equipment Layout; (2) Mod BAC-784MOD06 Revision of Emergency Floor Path Lighting; (3) Mod BAC-784MOD18 Revision of the PSU (overhead passenger service unit) installation. | 29.10.2012 |
| 10042007 | BOURNEMOUTH AVIATION (CONSULTANTS) | | | | 767-300 | In Flight Entertainment and In Seat Power Supply Revision | EASA.I.M.A.035 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | BAC Ltd Master Data List No. BAC-829MDL02, Issue 4 dated 29 October 2012. BAC Ltd Flight Manual Supplement No. BAC-829FMS02 Rev 1 for later revisions of the above listed documents approved by EASA. | The following BAC STC is a prerequisite for this modification: Mod BAC-784MOD02 | 30.10.2012 |
| 10016987 | BRISTOL AEROSPACE LTD. | EASA.I.M.R.S.01393 | | TCCA STC SH91-15; FAA STC SH26 | BELL 222 BELL 222B BELL 222U BELL 230 BELL 430 | Installation of Wire Strike Protection System. | | The installation is to be done in accordance with Bristol Aerospace Ltd. Drawing List EM103457 Issue E, DOT approved June 12, 2001 or later DOT approved Revision. | | 10.03.2008 |
| 10033724 | BRISTOL AEROSPACE LTD. | | | TCCA STC SH10-4 | 429 | Installation of Wire Strike Protection System. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | EM106490 Issue D dated 18 October 2010 - Modification Data Summary; PM-965-067 issue G dated 14 October 2010 - Instructions for Continued Airworthiness. | None. | 04.02.2011 |
| 10017097 | BRISTOW HELICOPTERS Ltd. | EASA.R.S.01020 | | | AS 332 L2 | Installation of FLIR Ultraforce II system | | Bristow Helicopters Modification MOD/332/25.5502 Bristow Helicopters RFM Supplement 3 Revision 1 or later approved revision | Restricted to Aircraft Serial Number 2616 G-REDN only | 27.07.2006 |
| 10017131 | BRISTOW HELICOPTERS Ltd. | EASA.R.S.01224 | REV. 1 | | AS 332 L, AS 332 L1 | Rockwell Collins TCAS II, TCAS 4000 System software upgrade from Version 7.0 to 7.1. Rev 1 covers modification of Rockwell Collins TCAS-4000 TCAS II update to version 7.1 software. Modification : MOD/332/34.5538 | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Bristow AS 332L/L1 Flight Manual Supplement No. 36-Rockwell Collins TCAS II, issue 4; - Bristow AS 332L/L1 Flight Manual Supplement No. 39-Rockwell Collins TCAS II Version 7.1, initial issue; or later revisions of the above listed documents approved b (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 06.08.2013 |
| 10017206 | BRISTOW HELICOPTERS Ltd. | EASA.R.S.01358 | | | S-92A | Installation of Additional Cargo Compartment Baggage Shelf. | | # FMS No 1, Initial Issue dated 19 July or later approved revision. # Maintenance Manual Addendum for Mod/S92/25.5643, Initial Issue dated 19 May 2007 or later approved revision. # Maintenance Manual Supplement for Mod/S92/25.5638, Initial Is (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 20.07.2007 |
| 10027446 | BRISTOW HELICOPTERS Ltd. | P-EASA.R.S.01370 | REV. 1 | | AS 332 L1 | ECT Night Vision Imaging System Installation Master Drawing NOR-0450 Rev. E. Note: STC revised due to new ownership Bristow Helicopters | As defined in TCDS EASA.R.002 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement AS332L-PMV-SUPP-002. Or later revisions of the above listed documents approved by EASA Master Drawing NOR-0450 Rev. E. Minimum Equipment List Supplement AS332L1-MEL-SUPP-002 Rev. B. Instructions for Continued Airworthiness NOR-046 (...) | None | 27.10.2011 |
| 10030289 | BRISTOW HELICOPTERS Ltd. | P-EASA.R.S.01472 | REV. 1 | | AS350B2 | System Configuration Modification No. MOD/350/00.5696 including installation of several avionics and electrical equipment items and miscellaneous equipment as specified in the Modification Sheet MOD/350/00.5696 Revision 1 to this STC introduces AS350B2 Series Flight Manual Change Sheet No. 15 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Change complies with BCAR Section G. The requirements for environmental protection and the (...) | Bristow Helicopters Ltd. AS350B2 Series Flight Manual Supplement No. 1 - Introduction of radio altimeter with dual indicator (Honeywell KRA-405B) - initial issue Bristow Helicopters Ltd. AS350B2 Series Flight Manual Change Sheet No. 6- Additional system (...) | Limited to S/Nos 1902 and 2312 | 06.09.2010 |
| 10034638 | BRISTOW HELICOPTERS Ltd. | EASA.R.S.01356 | REV. 1 | | S-76C | Qualification of the Garmin GNS-530 Series GPS for En-route Navigation and Non-precision approaches. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS29 Amendment 2 applicable to the change. The requirements for environmental protection and the (...) | Bristow Helicopters Limited Sikorsky S76 Flight Manual Supplement No 17, Issue 3, or later revisions of the above listed documents approved by EASA. Modification MOD/S76/34.5635, Issue 5, dated 21 December 2011. Certification Plan BHL/S76.4153 (>,<,> Issue d, dated 23 August 2011. | None. | 29.03.2012 |
| 10035156 | BRISTOW HELICOPTERS Ltd. | P-EASA.R.S.01356 | | | SIKORSKY S-76C | Qualification of the Garmin 400 Series GPS for enroute navigation and non-precision approach. | CS29 The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Bristow Helicopters Limited Sikorsky S76 Flight Manual Supplement No 20, Issue 1 or later revisions of the above listed documents approved by EASA. Modification MOQ/S76/34.5794 Issue 1 Certification Plan BHL/S76.4504 Issue A | None. | 30.05.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------|------------------|----------------|----------------------|---------------------|--|---|---|---|---------------|
| 10035471 | BRISTOW HELICOPTERS Ltd. | P-EASA.R.S.01467 | REV. 1 | | S-92A | Installation of Rockwell Collins TCAS II system Rev. 1 covers modification of Rockwell Collins TCAS-4000 TCAS II update to version 7.1 software - Modification S92/34.5702 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | S92 Flight Manual Supplement No 2 issue 3 - for software version 7.0 S92 Flight Manual Supplement No 4 initial issue - for software version 7.1 or later revisions of the above listed documents approved by EASA BHLS/92.0106 issue a dated 24 September 2010 (...) | None | 18.07.2012 |
| 10038558 | BRISTOW HELICOPTERS Ltd. | | REV. 1 | | S-76C | Modification of Rockwell Collins TCAS-4000 TCAS II update to version 7.1 software. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | S76 Flight Manual Supplement No 19 revision 1, S76 Flight Manual Supplement No 22 initial issue or later revisions of the above listed documents approved by EASA BHLS/76.4552 initial issue dated 29 March 2012 - Certification Plan; and MOD/S76/34.5784 (...) | Applicable only to the S76C with Turbomeca Arriel 2S2 engines. | 17.07.2012 |
| 10040678 | BRISTOW HELICOPTERS Ltd. | | | | S-61N | Activation of ADS-B Function using Mode S Extended Squitter | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS29 Amendment 2 The requirements for environmental protection and the associated certifi (...) | Bristow Helicopters Limited Sikorsky S61N Flight Manual Supplement No 30 Initial Issue or later revisions of the above listed documents approved by EASA Modification MOD/S61/34.5719 Issue 1 dated 23 May 2012 Certification Plan BHLS/61.5607 Initial issue dated 23 Mar 2012 | none | 18.07.2012 |
| 10045423 | BRISTOW HELICOPTERS Ltd. | P-EASA.R.S.01501 | | | EC 155 B1 | Activation of ADS-B Function using Mode S Extended Squitter | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 29 Amendment 3: CS 29.1301 CS 29.1309 CS 29.1431 CS 29.1581 The requirements for environmental protection and the associated (...) | Bristow Helicopters Limited Eurocopter EC155 Flight Manual Supplement No 1 Initial Issue or later revisions of the above listed documents approved by EASA Modification MOD/155/34.5720 Issue 1 Certification Plan BHLS/155.1075 Initial Issue | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 24.06.2013 |
| 10035883 | BRISTOW U.S. L.L.C. | | | FAA STC SR09577RC | S-76A, S-76C | Installation of an Automatic Deployable Emergency Transmitter (ADELT) | As per FAA TCDS H1NE The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-29 (Amdt.2) 29.610; Lightning Protection. The requirements for e (...) | Rotorcraft Flight Manual Supplement; RFMS-76-2009-01, Rev A or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) Master Drawing List; S76CPI-MDL-1_Rev (...) | none | 25.07.2011 |
| 10014176 | BRITISH AIRWAYS PLC | EASA.A.S.01312 | | | BOEING 747-400 | Omnibus Modification for B747 Mid J Project Stretch Embodiment | | British Airways Engineering Modification ref BA-747-25-0007 British Airways Engineering Master Data List ref. MDL-0049-06 Rev 00 or later approved revision. | | 28.02.2007 |
| 10014177 | BRITISH AIRWAYS PLC | EASA.A.S.01314 | | | BOEING 747-400 | Omnibus Modification for Boeing 747-436 Hi J Project Stretch Embodiment | | British Airways Modification BA-747-25-0008 | | 17.11.2006 |
| 10014178 | BRITISH AIRWAYS PLC | EASA.A.S.01315 | | | BOEING 777-200 | Omnibus Modification for B777 4 Class Project Stretch Embodiment. | | British Airways Master Data List Reference No. MDL-0055-08 Rev 00 dated 13 June 2008 or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 09.09.2008 |
| 10014376 | BRITISH AIRWAYS PLC | EASA.A.S.01651 | | | BOEING 737-400 | Installation of Enhanced Mode S | | British Airways Engineering Project Certification Plan Ref. CP-0002-04 Issue 1 or later approved revision. | None | 25.11.2005 |
| 10015033 | BRITISH AIRWAYS PLC | EASA.A.S.03345 | | | BOEING 757-200 | Reconfiguration of Passenger Cabin | | British Airways Engineering Master Data List MDL-0054-06 Issue 1, 20/05/2008 | | 13.06.2008 |
| 10015055 | BRITISH AIRWAYS PLC | EASA.A.S.03376 | | | BOEING 757-200 | B757 GPS upgrade to EGPWS. | | British Airways Master Data List MDL-0053-07 Issue 00, 12-12-2007. | BA B757 AFM Change Sheet No. 10, Page 1 and 3 (issue 2). | 28.03.2008 |
| 10028294 | BRITISH AIRWAYS PLC | EASA.A.S.01313 | | | BOEING 777-236 | | | | | 07.05.2009 |
| 10030714 | BRITISH AIRWAYS PLC | | | | B737-400 | Fuselage skin GPS antenna structural provisions | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-01- issue 2: Special Condition #Enhanced Airworthiness program for aeroplane Sys (...) | Master Data List - MDL-0070-10 Rev.00 dated 23 April 2010 (*) Drawing N°10098804 Rev.00 dated 23 April 2010 (*) Structure Design Report ESS-1711-PGB Rev.00 dated 20 April 2010 (*) Compliance check list ESS-1710-PGB Rev.00 dated 20 April 2010 (*) or I (...) | None | 02.07.2010 |
| 10031240 | BRITISH AIRWAYS PLC | | | | B737-400 | B737 GPS UPGRADE OF EGPWS | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-01- issue 2: Special Condition "Enhanced Airworthiness program for aeroplane Sys (...) | Master Data List MDL-0069-10 Rev. 00 dated 23/04/2010 (plus ED10105822)- Service Bulletin BA-737-34-0015 Rev. 00- AFM manual change request, BA Ref: 10098651 or later revisions of the above listed documents approved by EASA | None | 05.08.2010 |
| 10039421 | BRITISH AIRWAYS PLC | | | | A318/A319/A320/A321 | Airbus TCAS change 7.1 update. | In accordance with the TCDS EASA.A.064 Issue 7. The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions leve (...) | Master Data List, ref. doc. BA-320-34-0006-MDL, Revision 00, dated 15th March 2012- Compliance Check List, ref. doc. BA-320-343-0006-CCL, Revision 00, dated 2nd March 2012- Declaration of Compliance X440, dated 23rd April 2012 or later revisions of the above listed documents approved by EASA. | None. | 03.05.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|-------------------|----------------|----------------------|---|--|---|---|---|---------------|
| 10040257 | BRITISH AIRWAYS PLC | | | | 777-200 | B777 TCAS Change 7.1 update | CS25 Amendment 11The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcerti (...) | Master Data List BA-777-34-0003-MDL Issue 00, dated 20 March 2012,or later revisions of the above listed documents approved by EASA | none | 21.06.2012 |
| 10040970 | BRITISH AIRWAYS PLC | | | | B737-400 | B737 TCAS Change 7.1 Update. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS25 Amendment 11 | Master Data List BA-737-34-0016-MDL Issue 00 dated 01May12,or later revisions of the above listed documents approved byEASA. | None. | 09.08.2012 |
| 10041273 | BRITISH AIRWAYS PLC | | | | 747-400 | B747-400 TCAS change 7.1 update | The Certification Basis for the original product remains applicable tothis certificate/ approval.CS25 Amendment 11The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product are (...) | Master Data ListBA-747-34-0008-MDL Issue 00 dated 12Jun12,orlater revisions of the above listed documents approved by EASA | none | 03.09.2012 |
| 10043705 | BRITISH AIRWAYS PLC | | | | 767-300 | B767-336 TCAS change 7.1 update | The Certification Basis for the original product as amendedby thefollowing additional or alternative airworthiness requirements:CS 25 Amendment 12-CS 25.581CS 25.869CS 25.899CS 25.1301CS 25.1309CS 25.1316CS 25.1322CS 25.1351CS 25.1353CS 25.1357CS 25 (...) | Master Data ListBA-767-34-0009-MDL Issue 01or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noadverse effect upon the airworthiness ofthe produ (...) | 19.02.2013 |
| 10016535 | BRUCE AEROSPACE Inc. | EASA.IM.A.S.02912 | REV. 1 | FAA STC ST00941LA | BD-700-1A10 | Installation of Floor Proximity Emergency Escape Path Marking System. | EASA.IM.A.009The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product areunch (...) | EASA STC No. EASA.IM.A.S.02912 dated 15 September 2008.FAA STC No. ST00941LA dated 07 May 2008.Bruce Aerospace Master Drawing List No. 710-002 at the followingrevisions:-Rev G, Item 1, dated 1 May 2000-Rev L, Item 4, dated 7 July 2000-Rev AN, Item 2 (...) | Seat Mounted Proximity Lighting System:-Exit identifiers and flood lights (e.g. part number BR9279-series): aminimum oftwo out of five lamps in each part must be operative.-All light modules (e.g. part number BR9644-series): a minimum of oneout of two lamps in each part must be operative. | 30.06.2011 |
| 10035636 | BRUCE AEROSPACE Inc. | 2003-1912 | | FAA STC SA6011NM | CL-600-2B16 | Installation of Floor Proximity Emergency Escape Path Marking System | EASA.IM.A.023The Certification Basis for the original product remains applicable tothis certificate/ approvalThe requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product areunch (...) | Bruce Aerospace Master Drawing List CL600, Revision CE, Item 41, dated12 December 2006,or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions ofthis decision) | This system has been determined to comply with the applicableairworthiness requirements with the following operative lightassemblies:- Floor Mounted Proximity Lighting SystemExit identifiers and/or flood lights - 2 bulbs operative, light tracked mo (...) | 06.07.2011 |
| 10015035 | BUCHER LEICHTBAU AG | EASA.A.S.03348 | | | MD-11 | Modification of New Generation Stretcher (NGS) pn. 1120000-201 for installation in a MD-11 on the right hand side of the fuselage, in accordance with Bucher General Arrangement Drawing 1120000-201. | | CMM 25-67-015 Component Maintenance Manual NGS p/n 1120000-200/201 | This approval is only valid for the modified installation in the MD-11 on the right hand side of the fuselage.All otherinstallations of the New Generation Stretcher are covered by FOCA STC No. Z25-20-60It is the responsibility of the installer to ma (...) | 06.11.2007 |
| 10015074 | BUCHER LEICHTBAU AG | EASA.A.S.03404 | | FOCA STC: Z 25-20-53 | PILATUS PC-12 PILATUS PC-12/45 PILATUS PC-12/47 | EMS installation P/N 127001-300This EASA STC extends the validity of FOCA STC Z25-2053 for PC-12 S/N up to 888. | | AFMS 127001-300 Rev. C dated 3rd October 2007MDL MDOC-127001-300 Rev. C dated 3rd October 2007 | S/N 861-888 | 19.12.2007 |
| 10015092 | BUCHER LEICHTBAU AG | EASA.A.S.03429 | | | PILATUS PC-12 | Heavy Duty Floor Panel | | (A)FMS 0127470 Rev. N/C or ater CMM E016-19-08 Rev. N/C or later. | | 19.05.2008 |
| 10015227 | BUCHER LEICHTBAU AG | EASA.A.S.03644 | REV. 3 | | PILATUS PC-12/47E | Bucher EMS Equipment PC-12/47E,P/N 0127001-400: Addition ofequipment configuration P/N 0127001-450 | FAR 23 i.a.w. EASA TCDS A.089The Certification Basis for the original product remains applicable tothis certificate/ approval.No added safety valueThe certified noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | -Master Document List MDL E034-04-08 Rev. A (Config. P/N 0127001-400)-Aircraft Flight Manual Supplement FMS 0127001-400, NC (Config. P/N0127001-400)-Master Document List MDL E034-18-09 Rev. B (Config. P/N 0127001-450)-Aircraft Flight Manual Suppleme (...) | Applicable to PC12/47E series only.Smoking is not permitted in the aircraft cabin when the oxygen bottle is installed. | 07.07.2010 |
| 10016797 | BUCHER LEICHTBAU AG | EASA.IM.R.S.01150 | | | EUROCOPTER DEUTSCHLAND BO-105 | Ambulance Rescue SystemThe FOCA STC Z 25-20-49 (ambulance rescue system) was validated by the LBA under the number 0689/3025 dated 24 February 1995, approving installation onto the BO105SThis STC is re-issued as EASA STC to extend applicability to BO105LS. | | Flight Manual Supplement FMS-10-66 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 11.11.2005 |
| 10017089 | BUCHER LEICHTBAU AG | EASA.R.S.00697 | REV. 2 | | EUROCOPTER DEUTSCHLAND EC-135- | Ambulance and rescue equipment P/N ARA-EC135-AC65 in accordance with master document list MDOC-AC65 rev D or later approved revision.Revision 1: Change of Converter from ARA-EC135-AC65.Revision 2: Inclusion of models P2+ and T2+ | | # Rotorcraft Flight Manual Supplement No. FMS-AC65 rev c or later approved revision.# Master document list MDOC-AC65 rev D. | 1. No smoking is allowed in the rotorcraft cabin.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibil (...) | 12.01.2007 |
| 10017192 | BUCHER LEICHTBAU AG | EASA.R.S.01329 | REV. 1 | | EC135 P1, T1, P2, T2, P2+, T2+ | EMS-Equipment ARA-EC135-AC66Minor change to STC EASA.R.S.01329 to incorporate:- new mattress for the stretcher-additional stud to the stretcher installation-additional stud to the center cabinet- medical equipment retainers. | JAR-27 | -Master Document List E034-01-06, Rev. G-Rotorcraft Flight Manual Supplement FMS-AC6, Rev. C-Operational and Maintenance Manual E014-01-06 Rev. 2 | None | 24.07.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|-----------------------------|----------------|----------------------|---|--|---|--|---|---------------|
| 10017211 | BUCHER LEICHTBAU AG | EASA.R.S.01367 | | | EUROCOPTER DEUTSCHLAND EC-135- | EMS Installation ARA-EC135-AC61 This change introduces the following new items not included in FOCA STC FOCA 25-20-108: alternative side cabinet LH (p/n 1359270) new center cabinet (p/n 1359701) alternative seat installation (p/n 1353040). | | Flight Manual Supplement: FMS-AC61. Master Document List MDOC-AC6,1 Revision D Operation and Maintenance Manual: E014-04-04 | Smoking is not permitted in the rotorcraft cabin. | 29.05.2007 |
| 10017240 | BUCHER LEICHTBAU AG | EASA.R.S.01428 | | FOCA STC 25-20-62 | EUROCOPTER DEUTSCHLAND EC-135- | 1. Installation of ambulance and rescue equipment. 2. Extension of Validity of former FOCA STC Z - 25-2062 to the new model versions T2+P2+ of EC 135. | | 1. FMA-2 rev. E for Police Equipment EC135-AC62-POL, November 16, 2007. 2. MDL E034-06-07 Rev. A, dated November 19, 2007. | | 22.11.2007 |
| 10017275 | BUCHER LEICHTBAU AG | EASA.R.S.01509 | | FOCA STC 25-20-64 | EC-135-P1, EC-135-P2, EC-135-P2+, EC-135-T1 EC-135-T2, EC-135-T2+ | Installation of VIP-EC135-AC63 Bucher VIP Equipment (Revision of FOCA STC FOCA Z 25-20-64 Issue 3) | | (R)FM Supplement FMA-3 Rev. A or later approved revision-OMM VIP-EC135-AC63 Revision N/C- Master Document List MDOC-AC63 Revision D- FOCA STC Data sheet Z 25-20-64 Issue 3 or later revisions of the above listed documents approved by EASA | Not applicable | 14.11.2008 |
| 10027891 | BUCHER LEICHTBAU AG | FOCA 25-20-102 REV 3 | REV. 5 | | SEE AIRCRAFT ALLOCATION TABLE E043-01-12 REV. F DTD 25/09/12 | 16G New Generation Stretcher P/N 1120030 000 / -010 / -020 Minor Change to EASA STC 10027891 Rev. 4-Integration approval on A330-300 | JAR 25 Amdt 15, except § 25.562 as addressed per TGM/25/12 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List MDoc-1120030 Rev. L, dated 02/09/2013 Master Drawing List E015-01-3 Rev. C (>-<) dated 08/09/2013 Aircraft Allocation Table E043-01-12 Rev. F, dated 25/09/2012 Project Certification Plan E026-05-12 Rev. A, dated 02/09/2013 Statemen (...) | This STC constitutes an approval for installation only for aircraft serial numbers as stated in the Master Document List. For any additional configuration as per Aircraft Allocation Table, the requirements as stated in the Master Document List have to b (...) | 14.10.2013 |
| 10027884 | BUCHER LEICHTBAU AG | P-EASA.A.S.03796 | REV. 6 | | SEE AIRCRAFT ALLOCATION TABLE E043-04-12, REV. NC DATED 13/03/2013 | P/N 1120030-300 - 16g New Generation Stretcher XL Revision 3 to include 16G New Generation Stretcher P/N 1120030 -300 / -310 / -320 / -400 / -410 / -420-Introduction of ATR 42 type to the approved aircraft model list-Integration approval on ATR 42-500-1 (...) | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): H-01 - Instructions for Continued Airworthiness for Electrical Wiring Interconnection System (EWIS) JA (...) | Master Document List E034-03-08, Rev. E, dated 24/06/2013- Master Drawing List E015-01-09, Rev. B, dated 28/02/2013- Aircraft Allocation Table E043-04-12, Rev. NC, dated 13/03/2013- Project Certification Plan E026-14-12, Rev. B, dated 24/06/2013- User (...) | This STC constitutes an approval for installation only for aircraft serial numbers as stated in the Master Document List. For any additional configuration as per Aircraft Allocation Table, the requirements as stated in the Master Document List have to b (...) | 27.06.2013 |
| 10029211 | BUCHER LEICHTBAU AG | | REV. 5 | | EC135 | ARA-EC135-AC67 BUCHER Ambulance and Rescue System (ARA); Rev. 5 to EASA STC 10029211: - Installation of a hot compartment into center cabinet Record of revisions: Rev 4: editorial change to EASA approval certificate (correction by EASA) Rev 3: (Minor Cha (...) | JAR 27 The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Flight Manual Supplement: FMS-AC67; Rev. E- Airworthiness Limitation Section of Operation and Maintenance Manual: E014-03-09; Issue 04 or later revisions of the above listed documents approved by EASA- Master Document List E034-17-09 Revision I | Compliance with NVIS compatible flight decks must be demonstrated by the installer when ARA-EC135-AC67 with NVIS compatible cabin lighting shall be installed into NVIS equipped rotorcraft. | 21.05.2012 |
| 10030689 | BUCHER LEICHTBAU AG | P-EASA.A.S.03797 | | | SEE EASA APPROVED MODEL LIST | 16g New Generation Stretcher P/N 1120020-300 - (NGS) | JAR 25 Change 15 except 25.562 addressed per TGM/25/12 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- CRI H-01/Special Condition H-01 (...) | Master Document List ref. doc. No. E034-02-08 Rev. NC, dated 29th June 2010- User Manual ref. doc. No. E017-08-09 Rev. 02 or later revisions of the above listed documents approved by EASA | This STC does not constitute an approval for installation. Compliance with following requirements below need to be demonstrated through a design approval for Stretcher installation in aircraft per approved Model list: § 25.561(d) § 25.785(a) § 25.785(h) § (...) | 01.07.2010 |
| 10030758 | BUCHER LEICHTBAU AG | P-EASA.A.S.03712 | | | ERJ-170() | E006-14-08 - P/N 1120050 9G NGS XL Stretcher. Installation of removable 9g New Generation Stretcher (NGS) | | - Master Data List E015-09-08 Rev. A dtd. 5 June 2009 *) - User Manual E017-01-08 Rev. Initial dtd. 1 July 2009 *) or later EASA approved revision | - Limitations iaw. User Manual E017-01-08- TGM 25/12 addressed per CRI D-1 A.S.3712 | 13.07.2009 |
| 10030810 | BUCHER LEICHTBAU AG | FOCA Z25-20-57 | REV. 1 | | EC135 P1(CDS), EC135 P1(CPDS) EC135 P2(CPDS), EC135 P2+ EC135 T1, EC135 T2, EC135 T2+ | Installation of ARA-EC135-AC60 Ambulance and Rescue System New revision covers: STC issuance under EASA format at revision 0, Exchange of Suction Unit with new P/N 1355185 (>-<) adding P2+ and T2+ at the list of EC135 models. This STC revokes the STC 10030489. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement FMA-1 Rev. Cor later revisions of the above listed documents approved by EASA- Operation and Maintenance Manual E014-01-05 Rev. 00- Master Document List MDOC-AC60- Equipment List ARA-EC135-AC60 | None | 29.11.2011 |
| 10031757 | BUCHER LEICHTBAU AG | EASA.A.S.03984 | REV. 7 | | SEE AIRCRAFT ALLOCATION TABLE E043-02-12 REV. A DATED 10/12/2012 | ATS 16g Stretcher P/N: 1123000-000 / -100 / -200 / -300 / -390 / -400 / -500 / -590 Major Change to STC 10031757 Rev. 4-Installation envelope expansion for Boeing B787- Reduction of installation limitations- Introduction of the Aircraft Allocation Table | CS 25 Amendment 11 except § 25.562 addressed per TGM/25/12 The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Document List E034-01-08 Rev. K dated 10/12/2012- Master Drawing List E015-59-09 Rev. G dated 24/09/2012- Aircraft Allocation Table E043-02-12 Rev. A dated 10/12/2012- Project Certification Plan E026-11-12 Rev. A dated 10/12/2012- User Manual (...) | This STC constitutes an approval for installation only for aircraft serial numbers as stated in the Master Document List. For any additional configuration as per Aircraft Allocation Table, the requirements as stated in the Master Document List have to b (...) | 21.02.2013 |
| 10038839 | BUCHER LEICHTBAU AG | EASA.A.S.03919 | REV. 1 | | SEE EASA APPROVED MODEL LIST | Small person restraint system P/N 1120040 for use on 16G New Generation Stretcher P/N 1120020-XXX. | CS-25 The Certification Basis for the original product remains applicable to this certificate/ approval. | Refer to Master Document List No. E034-20-09 or later revisions of the above listed documents approved by EASA. | Small person restraint to be used only on stretcher P/N 1120020-XXX. Occupant limitations as per General Arrangement drawing 1120040 and User Manual E017-04-08. | 20.03.2012 |
| 10041271 | BUCHER LEICHTBAU AG | FOCA STC 25.20.42 (ISSUE 5) | REV. 1 | | SEE APPROVED MODEL LIST | ATL Airtransport Litter System P/N 123000 - Series including customer's options: P/N 123000-000 (Demo ATL) P/N 123000-110 (Lufthansa) P/N 123000-241 (AUA-2) P/N 123000-271 (Emirates-2) | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Component Maintenance Manual E016-10-96 Rev 01, replaced by Component Maintenance Manual E016-21-12 Rev 00, dated August 31, 2012 User Manual E017-01-94, Rev 02, replaced by User Manual E017-03-12 Rev 03, dated July 31, 2012 Master Document Register T (...) | 1. This STC does not constitute an approval for installation. Compliance with following requirements below need to be demonstrated by a design approval for Stretcher installation in aircraft per applicable approved model list: JAR 25.561(d) JAR 25.785(a), (...) | 15.03.2013 |
| 10014019 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.01073 | | | CESSNA 550 CITATION II | Installation of TCAS II system in accordance with MDL BCA 70-00 rev. 3 dated 18/03/05 | | - AFM supplement SMVBCA70 rev. 0 dated 18/01/05 | This STC is applicable to s/n 550-0622 | 04.04.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|----------------|----------------|----------------------|--|--|---|---|---|---------------|
| 10014024 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.01079 | | | BEECH 1900D | Emergency Transmitter Kannad 406 MHz installation on Beechcraft 1900 D. | | - BCA No. 61 Modification File.- BCA/RDO/140404 Drawing (Merely ELT installation).- BCA/RDO/030904 Drawing (ELT and Programming Unit). | 1. This STC is limited to Beechcraft 1900 s/n UE-1 and up.2. Required Pre-Modification: any existing EASA approved ELT. Doublet Airplane Flight Manual Supplement No. BCA/61 must be used with this modification. | 20.01.2005 |
| 10014049 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.01131 | REV. 1 | | BEECH 300 BEECH B300 | Installation of KLN 900 GPS with EFIS-85 according to STCEASA.A.S.01131. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 23 Amct. 23.62 The certificated noise and/ or emissions levels of the original pro (...) | File ref.: BCA 26 Rev 2 dated 03/10/2011 AFM SUPPLEMENT FOR GPS BENDIX / KING KLN-900 Ed. 01/12 - P/N BCA26 Rev2SMV. | BEECH 300 /B300 - F-GPRH - SN FA 226. | 06.02.2012 |
| 10014088 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.01182 | | | BEECH C90 BEECH C90 A | Installation of Weather Radar RDR-2000 | | Modification File No 50/BCA - rev 2 or later approved revision | Modification limited to Beech C90 S/N LJ-684 and Beech C90A/S/N LJ-1526 | 17.03.2005 |
| 10014121 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.01231 | | | BEECH C90 A | Skywatch SKY497 TCAS I installation on Beech C90A | | Modification file No 68/BCA revision 0 of 15th February 2005 | 1. AFMS No 68/BCA rev 0 of April 2005 must be used when modification is installed.2. Modification is limited to Beech C90A S/N LJ-1526 | 30.05.2005 |
| 10014190 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.01332 | REV. 1 | | CESSNA 500, CITATION CESSNA 550 CITATION II | Dual GARMIN GPS installation: GNS-430A + GNS-530A in accordance with MDL BCA 74-00 rev 6 dated 19/07/05 | | AFMS SMV BCA74 rev 7 dated 11/01/06 AMMS BCA74 rev 0 dated 07/07/05 | 1. S/N 550-0324 and 500-03612. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 31.01.2007 |
| 10014220 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.01381 | | | BEECH 1900D | Installation of ACS5 TCAS 2000 System. | | Doc. Ref. BCA 59/STC1248 dated April 2005, Revision 1. AFMS referenced BCA59 dated April 2005, Revision 1. | | 08.07.2005 |
| 10014335 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.01590 | | | BEECH 1900C, AIRLINER | Installation of ACS5 TAGS 2000 System on Beech 1900C. | | AFMS referenced BCA 55 dated April 2005 Revision 3. STC Document referenced BCA55/STC 1241 dated August 2004, Revision 1 | This STC is limited to Beech 1900 C SN UC-115 | 11.10.2005 |
| 10014387 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.01673 | | | BEECH 1900D | Installation of a TCAS II System | | AFMS Referenced BCA53 dated 30 May 2005 Aircraft Maintenance Manual Supplement BCA53-25 dated 23 June 2005 | This STC is applicable to Beechcraft 1900 model D, all Serial Numbers | 05.12.2005 |
| 10014616 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.02745 | | | BEECH 1900D | Installation of VIP configuration in aft ward cabin and passive noise reduction kit | | Airplane Flight Manual Supplement, Rev 0, dated 13 July 2006 or later approved revision. Airplane Maintenance Manual Supplement, Rev 0, dated 13 July 2006 or later approved revision. | This modification limits 14 passengers maximum. | 09.11.2006 |
| 10014831 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.03046 | | | BEECH C90 BEECH E90 BEECH F90 | Installation of dual Mode S transponder system (GARMIN GTX-330 or GTX-330D). | | Installation in accordance with BCA Master Drawing List BCA95-00, Rev 0, dated 03/02/2007, or later approved revisions. To be operated in accordance with BCA Flight Manual Supplement ref BCA95-05, Rev 1, dated 05/14/2007, or later approved revisions. (...) | | 22.05.2007 |
| 10014835 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.03051 | | EASA STC 2004-2474 | CESSNA 500 | Major Modification: change of a pilot HSI against EHSI - re-wiring of existing equipment power supply to the emergency bus. Note: This EASA STC supersedes EASA STC 2004-2474. | | Description: BCA35 Monopilot Citation I Rev. 8, AFM Supplement: SMV BCA35_Rev 5, Master Drawing List: BCA35-00_Rev. 5, BCA35-7_Etude des cas de panne_Rev. 1 | 1. See original STC : DGAC-F nr. F-CIM81-SF-000102. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatible (...) | 02.03.2007 |
| 10015004 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.03304 | REV. 1 | | FALCON 10 | Falcon 10 EHS Mode S Capability EASA.A.S.03304 has been reissued with reference 10015004. Rev. 1 following a transfer from ACI Systèmes to Business & Commuter Aircraft. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AFM Supplement ACI-05-AFM-M-048-Rev0 ACI-05-TD-M-048-Rev0 | Applicability for Falcon 10 that are already Mode A-(->->-) Can EL S Mode S certified and equipped with a dual transponder system. | 02.07.2013 |
| 10015029 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.03338 | | | BEECH 1900C, AIRLINER | Installation of a toilet cargo aera. | | Modification file BCA 103. | Modification limited to Beech 1900C s/n UC-168. | 10.01.2008 |
| 10015038 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.03351 | | | BEECH 1900 | 28-230 volts inverter installation on Beech 1900 series according to modification file BCA102, rev. 0, dated 30-10-2007 or later approved revisions. | | Airplane flight manual supplement BCA 102, rev. 0, dated 30-10-2007 or later approved revisions. | Airplane flight manual supplement BCA 102, rev. 0, dated 30-10-2007 or later approved revisions must be used when modification is applied. | 19.11.2007 |
| 10015058 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.03380 | | | BEECH 1900C, AIRLINER | Installation of a galley in the FWD right area of the A/C. | | 1. Installation must be in accordance with BCA 104 Master Data List Rev. 1, dated January 22, 2008 or later Approved Revisions; 2. Maintenance must be in accordance with BCA 104 Aircraft Maintenance Manual Rev. 1, dated January 22, 2008 or later Approved Revisions. | 1. Must be installed on Beech 1900C (S/N UC-1 and after); 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will intro (...) | 29.01.2008 |
| 10015059 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.03381 | REV. 1 | | BEECH 1900C, AIRLINER | Installation of a galley in the FWD right area of the A/C. | | Installation must be in accordance with BCA 106 Master Data List Rev. 1, dated January 22, 2008 or later Approved Revisions. Maintenance must be in accordance with BCA 106 Aircraft Maintenance Manual Rev. 1, dated January 22, 2008 or later Approved Revisions. | Must be installed on Beech 1900C (S/N UC-1 and after). | 29.01.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|----------------|----------------|----------------------|--|---|--|---|---|---------------|
| 10015111 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.03454 | REV. 1 | | BEECH BARON 58 | Dual Garmin Installation : GNS-530 (GPS approaches and PRNAV approved) + GNS 430 on Baron 58 (EFS 40/50 equipped). | | Master Drawing List Ref. BCA 107-00, revision 2, dated 23 April 2008. Airplane Flight Manual ref BCA107-06 Installation bulletin ref. 337, rev. 2 dated 22 February 2008 | Airplane Flight Manual Supplement ref. BCA 107-06, revision 2, dated 23 April 2008 or later approved revision must be used when modification is applied in the Aircraft Modification limited to s/n TH-1510, TH-1515, TH-1537, TH-1701 to TH-1706, TH-1734 and TH-1736. | 08.05.2008 |
| 10015140 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.03496 | | DGAC AAC NO. 751 | BEECH 1900C, AIRLINER | Installation of a King/ Bendix KLN-90B GPS with B-RNAV capacity. Note: This STC revises DGAC approval AAC No. 751. | | STC TRANSAIR No. 524 Revision 6, dated 09-01-2008. | 1. The supplemental AFM No. TF524PA must be used when the modification is installed. 2. Limited to S/N UC-168, F-GVLC and UC-153, F-GNPM. | 28.02.2008 |
| 10015232 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.03655 | | | BEECH 200 BEECH 300 BEECH B200 BEECH B300 | Be #Mode S / Flight ID# capable Collins TDR-94(D) transponders. | | z. The installation will be made in accordance with Business and Commuter Aircraft document BCA108-00, Revision 0, as of 14.08.2008 or later EASA approved revision. z. The Aircraft Flight Manual Supplement applicable to this modification is BCA108 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 27.04.2009 |
| 10015239 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.03666 | | | BEECH C90 BEECH C90A BEECH E90 BEECH F90 | BCA 110 Rev. 3 - Garmin GNS-x30WA (GNS-530WA or GNS-430WA) and Sandel SN4500 installation for B-RNAV, P-RNAV and GNSS approaches. | | AFM Supplement BCA110-07 rev 2, dated 28 November 2008. AMM Supplement BCA110-06 rev. 0, dated 22 August 2008. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 09.01.2009 |
| 10015291 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.03830 | REV. 1 | | 200C | BCA112 revision 2, dated 15 July 2012 : underwings, rooftop and POD antennas installation and belly antennas relocation. Major Change of STC EASA.A.S.03830. | The Certification Basis for the original product remains applicable to this certificate / approval. The certificated noise and/or emissions level of the original product are unchanged and remain applicable to this certificate / approval. | BCA112-04: Airplane Maintenance Manual Supplement, revision 2 dated 15 July 2012. BCA112-05: Airplane Flight Manual Supplement, revision 1 dated 19 March 2003, or later revisions of the above listed documents approved by EASA. | Limitation to sn BB-1473 | 25.07.2012 |
| 10015307 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.03899 | | | BEECH B90 BEECH C90 BEECH E90 | Be #Mode S / Flight ID# capable Collins TDR-94(D) transponders. | | z. The installation will be made in accordance with Business and Commuter Aircraft document BCA108-00, Revision 0, as of 14.08.2008 or later EASA approved revision. z. The Aircraft Flight Manual Supplement applicable to this modification is BCA108 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 27.04.2009 |
| 10015308 | BUSINESS & COMMUTER AIRCRAFT | EASA.A.S.03900 | | | BEECH F90 | Be #Mode S / Flight ID# capable Collins TDR-94(D) transponders. | | z. The installation will be made in accordance with Business and Commuter Aircraft document BCA108-00, Revision 0, as of 14.08.2008 or later EASA approved revision. z. The Aircraft Flight Manual Supplement applicable to this modification is BCA108 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 27.04.2009 |
| 10027323 | BUSINESS & COMMUTER AIRCRAFT | | | | B300 | HF-9000 installation on Beech B300 Series PROLINE 21 Equipped | FAR Part 23 and FAR Part 25- The Certification Basis for the original product remains applicable to this certificate- The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate | BCA 115 modification file or later revisions of the above listed documents approved by EASA | Modification applicable to B300 equipped with Proline 21, serial number FL-1 and after and FM-1 and after | 21.09.2009 |
| 10029562 | BUSINESS & COMMUTER AIRCRAFT | | | | BEECH C90A | BCA117 rev 0 - GPS replacement and aircraft capacity validation for B-RNAV, P-RNAV and GNSS approaches procedures | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the certificated noise and/or emissions levels | BCA117-06 rev 0 dated 10/12/2009 Airplane Flight Manual Supplement-BCA117-09 rev 0 dated 10/12/2009 Airplane Flight Maintenance Manual Supplement or later revisions of the above listed documents approved by EASA | Limited to: S/N LJ-1660 | 01.04.2010 |
| 10032469 | BUSINESS & COMMUTER AIRCRAFT | | | | C90 | BCA118-00 - ADF system upgrade on beech S/N LJ - 894 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | | Limitation to beech C90 SN LJ-894 | 09.11.2010 |
| 10034237 | BUSINESS & COMMUTER AIRCRAFT | | REV. 1 | | 1900D | Double GARMIN GPS400W Installation for B-RNAV and P-RNAV and RNP approach LNAV procedures, according to modification No BCA 121. Revision 1 adding the coupling to Rockwell Collins FYD-65 H flight director. | The Certification Basis for the original product remains applicable to this certificate. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AFM Supplement No. BCA-121-06 revision 1 dated 24/04/2012. AMM Supplement No. BCA-121-05 revision 0 dated 23/12/2010. Master Drawing List No. BCA-121-00 revision 1 dated 24/04/2012, or later revisions of the above listed documents approved by EASA. | Installation limited to Beech 1900D equipped with Rockwell Collins APC-65H autopilot or FYD-65 H flight director and EFIS Rockwell Collins DPU 84, upgraded for GPS approach through Rockwell Collins SB 77(523-0805196) and SB 81 (523-0805046). | 15.06.2012 |
| 10034238 | BUSINESS & COMMUTER AIRCRAFT | | | | 1900C, 1900D | BCA 120 revision 0 - 28-230 volts inverter(s) Installation on Beech 1900 Series. | The Certification Basis for the original product remains applicable to this certificate. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Airplane Flight Manual supplement BCA120-A revision 0 dated 09 March 2011. Airplane Flight Manual supplement BCA120-B revision 0 dated 09 March 2011 for later revisions of the above listed documents approved by EASA. | Not applicable | 17.03.2011 |
| 10034622 | BUSINESS & COMMUTER AIRCRAFT | | | | BEECH 1900D | BCA 85 rev 3 - Full VIP Cabin Installation Beech 1900D | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | None | None | 18.04.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|---------------------|----------------|-----------------------|---|--|---|--|--|---------------|
| 10035661 | BUSINESS & COMMUTER AIRCRAFT | | | | B200 | Installation of a ADF60 on a Beech 200 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | none | none | 08.07.2011 |
| 10035662 | BUSINESS & COMMUTER AIRCRAFT | | | | B200 | Installation of a ALT4000 on a Beech 200 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | none | none | 08.07.2011 |
| 10035665 | BUSINESS & COMMUTER AIRCRAFT | | | | B200 | Installation of a DME4000 on a Beech 200 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | none | none | 08.07.2011 |
| 10038717 | BUSINESS & COMMUTER AIRCRAFT | EASA, I.M.A.S.00327 | REV. 1 | | B1900D | Installation of Two Transponders mode S RCZ-852 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. FAR 23 Amdt 23-62 The certificated noise and/or emissions levels of the original product (...) | File ref.: BCA 60 Rev 1 dated 11/01/2012 Installation Manual RCZ-852 No. A 15-3851-001 | BEECH 1900 D S/N UE 70 | 09.03.2012 |
| 10043461 | BUSINESS & COMMUTER AIRCRAFT | | | | 200, 200T, B200C, B200T. | BCA 139 Revision 0b dated 28 Jan. 2013 - Rockwell Collins TCAS 4000 installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 23 Amdt 3 There are requirements for environmental protection and the associated certificate (...) | Airplane Flight Manual Supplement BCA 139-07 revision 0b dated 28 Jan. 2013 or later revisions of the above listed documents approved by EASA. | Prior to installation of this modification it must be determined if compatibility with transponder. Prior to installation of this modification it must be determined that the aircraft total load electrical consumption does not exceed 80% of the total ai (...) | 30.01.2013 |
| 10043505 | BUSINESS & COMMUTER AIRCRAFT | | | | 1. 200, 200T 2. B200, B200C, B200T | BCA 136 Revision 0 dated 28 Jan 2013 - Flight Display Systems FD200 CPU-8and FD102 CV Moving Map cabin installation. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Airplane Flight Manual Supplement BCA 136-02 Rev 0 dated 28 Jan 2013 or later revisions of the above listed documents approved by EASA | Installation limited to Proline 21 equipped aeroplanes. Prior to installation of this modification it must be determined that the aircraft total load electrical consumption does not exceed 80% of the total aircraft electrical capability. Prior to installation of this design change it must be determined. | 04.02.2013 |
| 10016240 | BUTLER NATIONAL CORPORATION | EASA, I.M.A.S.02512 | | FAA STC NO. ST00846SE | BOEING 747-100 BOEING 747-200 BOEING 747-300 | Installation of Model 840037-108, 840037-112, and 840037-113 transient Suppression Device on Boeing 747-100, 747-200, 747-300 SP and SR series airplanes fitted with Honeywell Analog Fuel quantity Indication System in accordance with Butler National C (...) | | | This STC is approved only for the product configuration as defined in the approved design data. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 15.10.2007 |
| 10016460 | BUTTERFIELD INDUSTRIES, INC. | EASA, I.M.A.S.02806 | | FAA STC SA4362NM | BEECH 200 BEECH B200 SERIES | Installation of Butterfield Industries Oil Residue Collection System Kit P/N BI 1011-1 | | Installation Data Package BI 1013, Revision B, dated November 12, 2007 or later approved FAA versions Drawing List BI 1012, Revision C, dated November 12, 2007 or later approved FAA versions | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 08.08.2008 |
| 10016461 | BUTTERFIELD INDUSTRIES, INC. | EASA, I.M.A.S.02807 | | FAA STC SA00829CH | BEECH B200 SERIES | Installation of Butterfield Industries Air/Oil Separator System Kit P/N BI 6001-1 | | Installation Data Package BI 6003, Revision H, dated November 12, 2007 or later approved FAA versions Drawing List BI 6004, Revision H, dated November 12, 2007 or later approved FAA versions | Aircraft which are not modified by STC SA4362NM, will also require installation of an oil collection tank 1 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any or (...) | 08.08.2008 |
| 10039322 | BUTTERFIELD INDUSTRIES, INC. | | | FAA STC SA02215SE | 1. F90 2. 200, 200C, 200CT, 200T A200, A200C, A200CT, B200CT 2. B200T, 300 3. B200, B200C, B300 | Inspection Panel Kit BI 9501-001. Fabrication of Inspection Panel Kit BI 9501-001, in accordance with Drawing List BI 9025, Revision A, dated February 17, 2012, or later FAA-approved revision; and installation in accordance with Installation Instructions (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. As established by FAA STC SA02215SE. The requirements for environmental protection and (...) | Drawing List BI 9025, Revision A, dated February 17, 2012 <(-,-) Installation Instructions BI 9020, Revision A, dated February 17, 2012 <(-,-) or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 25.04.2012 |
| 10015347 | C & D INTERIORS | EASA, I.M.A.S.00475 | REV. 1 | | BOEING 737-600 | C&D Interiors Storage S1, Galley S1, Galley G2B, Galley G3B, Storage S2, Forward lavatory, Passenger Seat Installation, Overhead Storage Bin Installation, Sidewall Installation, PSU Installation, Passenger Seat Proximity Light Installation. | | C&D Interiors Index List A81 Revision B, dated August 28, 2003 or later FAA-approved revision. | 1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until superseded, revoked or otherwise terminated. 2. If transfer of this STC Certificate is requested under Part 21A.116, the STC will be (...) | 17.05.2006 |
| 10015324 | C & D ZODIAC, Inc. | EASA, I.M.A.S.0028 | REV. 3 | FAA STC ST01415LA | CL-600-2C10 CL-600-2D24 CL-600-2E25 | Flight Deck Access System (FDAS) Installation | EASA IM.A.023 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRJ-H-01 The requirements for environmental protection and the associated (...) | FAA STC No. ST01415LA dated 21 December 2010-FAA SoC ref. 129L-10-85 dated 22 February 2010-C <(-&(-> D Zodiac MDL 9391000, Revision W, dated 26 October, 2010-C & D Interiors AFMSupplement 939-1-AFMS-002 or later revisions of the above listed documents (...) | None | 04.01.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|---------------------|----------------|----------------------|--|--|---|--|--|---------------|
| 10035678 | CALEDONIAN AIRBORNE SYSTEMS LTD | | | | JETSTREAM 4100 | Installation of Role Equipment for Maritime Patrol Training | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI C-02, Installation of Standard Passenger Seats in a Reconfigured Cabin Layout. The c (...) | Caledonian Airborne Systems Compliance Document and Accomplishment Instructions, F005/222/001, Issue F dated 7th July 2011, Caledonian Airborne Systems Flight Manual Supplement F014/222/001 Issue B dated 14th June 2011 or later revisions of the above listed documents approved by EASA | Serial numbers 41022 and 41025 only. | 11.07.2011 |
| 10034473 | CALIFORNIA SHOCK TRAUMA AIR RESCUE | | | FAA STC SR02160LA | BO 105 LS A-3 | Engine Upgrade: Removal of two (2) Rolls-Royce engine models 250-C28C and installing two (2) Rolls-Royce engine models 250-C30P in accordance with FAA STCSR02160LA, dated March 11, 2011. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | CALSTAR FAA approved Rotorcraft Flight Manual Supplement, CAL204, Revision I/R, dated July 30, 2008 or later revisions of the above listed documents approved by EASA - CALSTAR Master Drawing List CAL201, Revision B, dated May 16, 2008 - Installation o (...) | None. | 11.04.2011 |
| 10015309 | CAMERON BALLOONS LIMITED | EASA.BA.S.01001 | | | GEFA AS 105 GD (HOT AIR AIRSHI | Introduction of Liquid Pilot Lights. | | FMS-C474-STC-Iss 1 | None. | 30.04.2007 |
| 10016193 | CANARD AEROSPACE CORPORATION | EASA.IM.A.S.02430 | | FAA STC ST01106CH | BOEING 747-100 BOEING 747-100B SUD BOEING 747-200B BOEING 747-200C BOEING 747-200F BOEING 747-300 BOEING 747-SP BOEING 747-SR | Installation of Digital Flight Data Recorder Expanded Parameters in accordance with Canard Aerospace Corporation Document Control Drawing No. 30010, Rev. G, dated 6 October 2003, or later FAA approved revisions, FAA STC number ST01106CH. | | Data package containing the following documents, and additional supporting data: 1. Certification Plan 31-3314-01 Rev. B2. Top level Documents Control Drawing No. 30010 Rev. G, for 747-100 Series; 747-100B series, 747-100B SUD series; 747-200B Series; (...) | This STC is approved only for the product configuration as defined in the approved design data. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 29.10.2007 |
| 10016546 | CANARD AEROSPACE CORPORATION | EASA.IM.A.S.02925 | | FAA STC ST01117CH | BOEING 747-100 BOEING 747-100B SUD BOEING 747-200 BOEING 747-200C BOEING 747-200F BOEING 747-300 BOEING 747-SP BOEING 747-SR | Option 1: Installation of Honeywell Classic Navigator system in accordance with Canard Aerospace Corporation Control Drawing Nr. 30011, Rev. NC, dated November 25, 2002, Appendix A, or later FAA approved revision. Option 2: Installation of a CMA-900 (...) | | Option 1: 34-3249-04-NC34-3249-09-NC Option 2: 34-3275-04-B 34-3275-08-E 34-3275-10-NC Option 3: 34-3404-06-NC34-3404-07-NC Airplane Flight Manual Supplement Airplane Maintenance Manual Supplement Airplane Flight Manual Supplement Instruction for Continue (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 12.11.2008 |
| 10028374 | CANARD AEROSPACE CORPORATION | EASA.IM.A.S.03246 | REV. 1 | FAA STC ST01331CH | DHC-8-201, DHC-8-202 | Installation of Honeywell Mark VII Enhanced Ground Proximity Warning System to replace the existing Mark II Ground Proximity Warning System | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation must be in accordance with Canard Aerospace Document Control Drawing No. 40094, Revision O, dated December 12th, 2012 Operation must be performed in accordance with FAA approved Airplane Flight Manual Supplement, Canard Aerospace No. 34-3250 (...) | Electronic Flight Information Systems (EFIS) symbol generator upgraded to Part Number (PIN) 7004544.314 in accordance with Honeywell Service Bulletin (SB) 7004544.34-48 (modification BC) is required prior to, or concurrent with, accomplishment of this (...) | 26.03.2013 |
| 10014684 | CARGOLUX AIRLINES INTERNATIONAL | EASA.A.S.02847 | | | BOEING 747-400 SERIES | Installation of an Artex B406-4 Automatic Fixed Emergency Locator Transmitter on B747-400F in accordance with Cargolux Service Bulletin CLX-SB-747-23-001 and Master Drawing list CLX-STC-06-01. | | The publications affected as tabled in the Service Bulletin shall be updated using the information provided in report CLX-STC-06-01 Paragraph 7.4, Publication Update | | 14.03.2007 |
| 10015049 | CARGOLUX AIRLINES INTERNATIONAL | EASA.A.S.03366 | | | BOEING 747-400F | ICS-220A Iridium satcom installation on B747-400F airplanes in accordance with Master Drawing List CLX-STC-07-01, dated 22 November 2007 or later EASA approved revisions. | | Data package containing the following documents and additional supporting data: Engineering Data Rept CLX-STC-07-01, dated 22 November 2007. Master Drawing List CLX-STC-07-01. Drawing CLX-STC-07-01. Boeing SSM and WDM for Satcom and AMU | The Data on Airworthiness Limitations, revisions required of the Maintenance publications etc. arising from the changes associated with the STC is provided in DWG CLX-STC-07-01: ICS-220A Iridium Satcom Installation on B747-400F Airplanes - ICA and miscellaneous information. | 22.04.2008 |
| 10028509 | CARSON HELICOPTERS, INC. | P-EASA.IM.R.S.01532 | | FAA STC SR02487NY | SIKORSKY S61N AND S61NM | Increased Category A Performance. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 29 at Am 35. | Modification Definition: Carson Helicopters Inc. Certification Plan for S61N Increased Category A Performance when equipped with Carson Composite Main Blades - Ref: Report CHI-04-1000 Rev B Dated 08/29/2006 Flight Manual Supplements Carson Helicopters Inc (...) | As specified in the approved RFMS. | 18.01.2010 |
| 10032448 | CARSON HELICOPTERS, INC. | EASA.IM.R.S.01138 | | FAA STC SH639NE | S61 N | Installation of 11,000 pound capacity Cargo Sling in accordance with Carson Services, Inc., Drawing List Report No S61N-1A Revision D dated January 10, 2005 | | - Carson Services, Inc. FAA approved Flight Manual Supplement No 1, as approved February 2, 2001 (or later approved revision) to FAA approved Sikorsky Flight Manual Supplement No. 15 - Carson Services, Inc. FAA approved Flight Manual Supplement No. 1, asa (...) | None | 06.07.2009 |
| 10043050 | CARSON HELICOPTERS, INC. | EASA.IM.R.S.01129 | | FAA STC SR01585NY | SIKORSKY S-61N SIKORSKY S-61NM | Installation of Carson Helicopters Inc. composite main rotor blades, p/n 163-101-1 as replacement for the Sikorsky metal blade and with performance improvement: i.e. performance equal to that of the Sikorsky basic RFM and associated Sikorsky RFM Supplements 15 and 16. | | - Instructions for Continued Airworthiness for CHI Composite Main Rotor Blade, Part number 163-101-1. Report CH-MR-ICA Rev 0 dated Jan 13th 2003 or later approved revision. - CHI Installation Instructions CSI install-I Rev A dated July 23rd 2002 or late (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The (...) | 08.06.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|---------------------------------|----------------|--------------------------------|--|--|---------------------|--|---|---------------|
| 10015717 | CASCADE AEROSPACE Inc. | EASA.IM.A.S.01714 | | SA00-69 | BOMBARDIER DHC-8-100 SERIES BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-300 SERIES | Environmental Control System Performance Improvement Modification as per TCCA STC SA00-69, Issue 2, Issue Date: 15 June 2000 | | For configuration 1:1. Conair Aerospace Maintenance Manual Supplement MMS 2302-1 dated 09 June 2000*For configuration 2:1. Conair Aerospace Maintenance Manual Supplement MMS2302-1 dated 09 June 2000*2. Conair Aerospace Maintenance Manual Su (...) | Configuration 1:a) Fwd Exhaust per Conair Aerospace Drawing 2302D21003 Rev Nil, dated May 19, 2000*.b) Aft Exhaust per Conair Aerospace Drawing 2302D21005 Rev Nil, dated May 19, 2000*. This element applicable to -300 Series only.c) Fwd Intake per Conair Aerospace Drawing 2302D21002 Rev Nil. | 22.02.2006 |
| 10015908 | CASCADE AEROSPACE Inc. | EASA.IM.A.S.02035 REVISION 1 | | TCCA STC SA06-58 | BOEING 747-400 | Design Elementary & Enhanced Surveillance.Validation of a STC SA06-58. | | Data package containing the following documents and additional supporting data:1. Certification Plan Boeing 747-400 Series Elementary & Enhanced Surveillance Program. Cascade Aerospace Engineering document PR 1-3710.1.1, Rev. Nil, or later approv (...) | | 20.11.2006 |
| 10015925 | CASCADE AEROSPACE Inc. | EASA.IM.A.S.02060 | | SA01-98 | BOMBARDIER DHC-8-102 BOMBARDIER DHC-8-103 BOMBARDIER DHC-8-311 BOMBARDIER DHC-8-315 | APU Oil Cooler Door Deletions per TCCA STC SA01-98, Issue 1, Issue Date: 23 August 2001 | | Conair Aerospace Engineering Order EO 2977, approved on 23.August 2001 or later Transport Canada approved revisionConair Aerospace Maintenance Manual Supplement MMS 2977-1 dated 27. June 2001 or later Transport Canada accepted revisionConair Aerospa (...) | Configuration 1:DHC-8 Series 100/200/300 aircraft modified in accordance withSTC EASA.IM. A.S. 01714Configuration 2:DHC-8 Series 100/200/300 aircraft in accordance with STC EASA.IM.A.S 02061 (equivalent to Bombardier Mod 8/2328 and 8/3330 Conditions (...) | 07.08.2006 |
| 10015926 | CASCADE AEROSPACE Inc. | EASA.IM.A.S.02061 | | SA01-67 | BOMBARDIER DHC-8-102 BOMBARDIER DHC-8-103 BOMBARDIER DHC-8-311 BOMBARDIER DHC-8-315 | Screen Door and Ram Air Vent Modifications per TCCA STC SA01-67, Issue 1, Issue Date: 7. June 2001 | | Conair Aerospace Engineering Order EO 2979-1, Transport Canada approved on 7. June 2001 or later Transport Canada approved revision)Conair Aerospace Maintenance Manual Supplement MMS 2979-1 dated 27 April 2001 or later Transport Canada accepted revision) | Configuration 1:DHC-8 Series 100/200/300 aircraft with or without Bombardier Standard option Order 8155 APUConfiguration 2:DHC-8 Series 100/200/300 aircraft with or without Bombardier ACM Exhaust Mod. 8/1815Configuration 3:DHC-8 Series 100/200/300 a (...) | 07.08.2006 |
| 10016096 | CASCADE AEROSPACE Inc. | EASA.IM.A.S.02285 | | TCCA STC SA07-44 | BOMBARDIER CL-600-2B19 | Conversion of full cabin class E Cargo Compartment. | | ¿ Cascade Aerospace Inc. Engineering Master Document List EMDL 3683, Revision 5, dated 03 April 2007 or later approved revision.¿ Cascade Aerospace Inc. Flight Manual Supplement FMS 3683, Revision IR, dated 03 April 2007 or later approved revi (...) | | 11.04.2007 |
| 10016357 | CASCADE AEROSPACE Inc. | EASA.IM.A.S.02679 | | TCCA STC SA 08-08 ISSUE 1, MAR | BOMBARDIER DHC-8-400 SERIES | Forward Baggage Compartment Installation. | | Engineering Order EO 4049, Rev. IR, dated 14 February 2008Master Document List MDL 4049, Rev IR, dated 14 February 2008Maintenance Manual Supplement MMS 4049 Rev. IR, dated 14, February 2008Maintenance Requirements Manual Supplement MRMS 4049, Rev. IR, dated February 14, 2008 | | 12.03.2008 |
| 10016358 | CASCADE AEROSPACE Inc. | EASA.IM.A.S.02680 | | TCCA STC SA08-9, ISSUE 1, 4 FE | BOMBARDIER DHC-8-400 SERIES | G4 Galley Installation. | | Engineering Order EO 4050, Rev IR, dated January 23, 2008Master Document List MDL 4050, Rev IR, dated January 2, 2008Maintenance Manual Supplement MMS 4050, Rev IR, dated January 23, 2008Maintenance Requirements Manual Supplement MRMS 4050, Rev IR, dated January 23, 2008. | | 08.02.2008 |
| 10016359 | CASCADE AEROSPACE Inc. | EASA.IM.A.S.02681 | | TCCA STC SA08-10, ISSUE 1, 4 F | BOMBARDIER DHC-8-400 SERIES | Right Aft Draft Bulkhead InstallationTCCA STC Number: SA08-10, Issue 1, 04 February 2008 | | Engineering Order EO 4051, Rev IR, dated January 23, 2008.Master document list MDL 4051 Rev IR, dated January 23, 2008Maintenance Manual Supplement MMS 4051, Rev IR, dated January 23, 2008Maintenance Requirements Manual Supplement MRMS 4051, Rev IR dated January 23, 2008 | | 08.02.2008 |
| 10016360 | CASCADE AEROSPACE Inc. | EASA.IM.A.S.02682 | | TCCA STC SA 08-11 ISSUE 1 4 FE | BOMBARDIER DHC-8-400 SERIES | 72 Passenger Seat Reconfiguration | | Engineering Order EO 4052, Rev IR, dated January 23, 2008.Master Document List MDL 4052 Rev IR, dated January 23, 2008Maintenance Requirements Manual Supplement MRMS 4052, Rev IR,dated January 23, 2008. | | 08.02.2008 |
| 10016361 | CASCADE AEROSPACE Inc. | EASA.IM.A.S.02683 | | TCCA STC SA 08-12 ISSUE 1 4 FE | BOMBARDIER DHC-8-400 SERIES | G3 Galley Modification. | | Engineering Order EO 4053, Rev IR, dated January 23, 2008.Master Document List MDL 4053, Rev IR, dated January 23, 2008Maintenance Manual Supplement MMS 4053, Rev IR Dated January 23, 2008Maintenance requirements Manual Supplement MRMS 4053, Rev IR dated January 23, 2008. | | 08.02.2008 |
| 10016563 | CASCADE AEROSPACE Inc. | EASA.IM.A.S.02954 | | TCCA STC NUMBER SA08-91 | BOMBARDIER DHC-8-400 SERIES | DHC-8-400 Conversion to Bulk FreighterTCCA STC Number SA08-91, Issue 1, dated 31 December 2008 | | Installation of the freighter conversion must be in accordance with Cascade Aerospace Inc. EO 4079, Rev. 2, TCCA approved 12 December 2008 and Master Document List MDL 4079, Rev.2, TCCA approved 24 December 2008 *.Operation of the aircraft must be (...) | 1 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 29.01.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|----------------------|----------------|------------------------|---|--|--|--|--|---------------|
| 10029545 | CASCADE AEROSPACE Inc. | | | TCCA STC SA01-40 | DHC-8-102, -103, -106 DHC-8-201, -202 DHC-8-3-1, -302 | Ground Air Connection Modification TCCA STC SA01-40, Is.1, dated 14 March 2001 | FAR 25 as per EASA.I.M.A.191 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product remain applicable to this certificate/ approval. | - Installation of Air Ground Connection Modification is to be carried out in accordance with Conair Aerospace Engineering Order EO 2346, Rev. 1, dated 01 August 2002. - Maintenance is to be carried out in accordance with Conair Aerospace Engineering Maintenance (...) | Placards have to be installed in accordance with Conair Aerospace Engineering drawing 2854D21501, Rev. D. | 31.03.2010 |
| 10042988 | CASCADE AEROSPACE Inc. | EASA.I.M.A.S.02035 | REV. 1 | TCCA STC SA06-58 | B747-400 | Design Elementary & Enhanced Surveillance. Validation of a STC SA06-58. | | Data package containing the following documents and additional supporting data: 1. Certification Plan Boeing 747-400 Series Elementary & Enhanced Surveillance Program, Cascade Aerospace Engineering document PR1-3710.1.1, Rev. Nil, or later approved rev (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 20.11.2006 |
| 10015469 | CASCADE AVIATION SERVICES INC. | EASA.I.M.A.S.01222 | | SA01513SE | BAE JETSTREAM 3201 SERIES | Installation of Sandel Terrain Awareness Warning System (TAW/S) | | FAA STC SA01513SEMDL: 04CAS328-MDL, Rev DAFM: 04CAS328-D80, Rev C | Limitations as per Flight Manual | 09.02.2007 |
| 10015711 | CASCADE AVIATION SERVICES INC. | EASA.I.M.A.S.01706 | REV 1 | FAA STC ST01630SE | BOEING 737 | Installation of Eyebrow Window Deletion Kit. FAA STC ST01630SE | | Master Drawing List 05CAS469-MDL, revision A, dated December 8th, 2005, or later EASA approved revision | Compatibility of this design change with other previously approved modifications must be determined by the installer. | 14.03.2006 |
| 10016109 | CASCADE AVIATION SERVICES INC. | EASA.I.M.A.S.02299 | | STC ST01181SE | DORNIER 328-100 DORNIER 328-300 | Installation of increased security cockpit door | | Cascade Aviation, Inc. Master Drawing List (MDL) 02CAS096-MDL, Revision D, dated July 14, 2003 or later approved revision | A copy of this Certificate and Cascade Aviation Services Master Drawing List (MDL) 02CAS096-MDL, Revision D, dated July 14, 2003 or later approved revision must be maintained as part of the permanent records for the modified aircraft | 19.01.2007 |
| 10027057 | CASCADE AVIATION SERVICES INC. | P-EASA.I.M.A.S.03259 | | FAA STC ST01133SE | DHC-8-102, -103, -106 DHC-8-201 DHC-8-301, -311, 314, 315 | Installation of an increased security cockpit door. FAA STC ST01133SE | FAR Part 25 as per TCDS EASA.I.M.A.191 for DHC-8-100/200/300 Series | Installation must be performed in accordance with Cascade Aviation Services, Inc. Master Drawing List 01CAS029-MDL, Rev. H, dated 16.02.2006. Operation must be performed in accordance with AFM Supplement Doc. No. 01CAS029-D 80. Maintenance must be performed (...) | None | 02.09.2009 |
| 10016467 | CAV AEROSPACE Inc. | EASA.I.M.A.S.02813 | | FAA STC SA01435WI | LC41-550FG (C400) | Installation of TKS Ice Protection System Components. Model LC41-550FG (C400) | | According to AS&T or CAV Master Drawing List No. 12200-DL, Rev. 2 or later FAA approved Revisions To be operated and maintained in accordance with FAA approved AS&T or CAV AFM-S Document No. 12200-FMS, original issue, or later FAA approved Revisions | 1. Prohibited for operations in known icing conditions. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce (...) | 24.04.2009 |
| 10048108 | CAV AEROSPACE Inc. | | | | A. A36 (SERIAL E-2112 & AFTER) B. G36 | Installation of TKS fluid Ice Protection System for flight into known icing conditions. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Aerospace Systems & Technologies (AS-><->T) Drawing List No 10300SL, Revision 3 dated February 25, 2003. Operation in accordance with Supplement to Pilot's Operating Handbook and/ or FAA Approved Flight Manual, Document (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce adverse effect upon the airworthiness of the product. | 14.08.2013 |
| 10039180 | CEAPR | | REV. 2 | | HR 200/100, HR200/100S HR200/120, HR200/120B HR200/160 R2100, R2100A, R2112, R2120U R2160, R2160D, R2160I | DET 120301 Rev. 0 - Oil pressure transducer line R2000-HR2000 dated 7/March 2012 | The certification basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 23CS 23 amdt 2 The certificated noise and/ or emissions level of the original product (...) | None | Limitation to serial numbers 001 up to 378 included. | 26.04.2012 |
| 10032548 | CENTURY FLIGHT SYSTEMS, INC | | | AK 847 STC SA 3389SW-D | PA-31, PA-31-325, PA-31-350 | AK 847 Validation / Century 41 Installation in Piper, Type Certificate Number A20SD models PA-31, PA-31-325, PA-31-350 | The Certification Basis for the original product remains applicable to this certificate/ approval. No additional airworthiness requirements or special conditions are associated with this change in accordance with Annex to 1702/2003 21A.101(b) and associa (...) | 68S803 Pilot's Operating Handbook, January 1985 - Original FAA Approved Supplement to Pilot's Operating Handbook and/ or FAA Approved Airplane Flight Manual Supplement, P/N 68S926, Revision 1, dated 8-5-81 is required for Piper Models PA-31 S/N712 and Up (...) | 1. This STC is limited to aircraft Models PA-31, PA-31-325 and PA-31-350 (<-> <-> 2. Piper Model PA-31 is limited to S/N 712 and Up (<-> <-> 3. Placard 13A990-1 stating "conduct trim check prior to first flight of day (see POH)" is required | 15.11.2010 |
| 10015954 | CERTIFICATION DOCUMENTS AND | EASA.I.M.A.S.02097 | | FAA STC ST10674SC | CESSNA 750 CITATION X | Installation of an Aerodata AG Automatic Flight Inspection System (AFIS). | | Installation in accordance with Master Drawing List FISR00002, Rev A, dated June 5, 2007, or later approved revision. Operation to be in accordance with Approved Airplane Flight Manual Supplement ref. AFISR00011, Rev IR, dated June 28, 2007, or later (...) | Limited to Aircraft Serial Number 262 only. | 15.10.2007 |
| 10043042 | CERTIFIED TECHNOLOGY | P-EASA.I.M.R.S.01538 | | FAA STC SR00409SE | 368D, 369E, 369FF 369H, 500N, 600N | Rapid Door Removal System Hinges. | | - Certification Plan and Regulation Checklist MASOO-001, rev. NC, dated April 16, 2009. - Drawing List CT9602, rev. H, dated April 30, 2009. - Installation Instructions CT9602-1, rev. E, dated January 9, 2009. - Instructions for Continued Airworthiness CT9602-2, rev. -, dated April 16, 2009. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The (...) | 16.06.2009 |
| 10014339 | ČESKA LETECKÁ SERVISNÍ A.S. | EASA.A.S.01597 | | | MITSUBISHI MU-2B-36A | Avionics Upgrade in accordance with doc No CLS-05-MU-AC128, Revision 1.3., dated 27/01/2006. | | Mitsubishi MU2-B Flight Manual Supplement No AC128-2501, Rev. 3 dated 27/01/2006 Mitsubishi MU2-B Maintenance Supplement, No AC128-2701, Rev. 1.1 dated 27/01/2006 | Limited to S/N MU2B-36A-7185A. | 08.02.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|---|---|--|---|--|---------------|
| 10014486 | CESKA LETECKA SERVISNI A.S. | EASA.A.S.02548 | | | LETECKE ZAVODY L 200 D | Avionics upgrade in accordance with the document no.CLS-06-L200-AC 129, dated 23/06/2006 | | L 200 D Flight Manual Supplement, No Do-L200D-1011.1, dated June 2006 | S/N 171206 | 28.07.2006 |
| 10014761 | CESKA LETECKA SERVISNI A.S. | EASA.A.S.02951 | | | CESSNA 421C | Avionics Upgrade, Prj. No.: CLS-07-C421-AC147/Replacement of existing obsolete transponder, altimeters, ELT and COM/NAV with new that meets the current technical and operating requirements. | | - Supplement to Flight Manual of Cessna 421C; Pilot's operating handbook supplement, issue August, 2007 | | 15.10.2007 |
| 10014901 | CESKA LETECKA SERVISNI A.S. | EASA.A.S.03135 | | | CESSNA 172S | Installation of the DME KN 62A, ADF KR-87 and XPDR KT-73 | | CSL-07-C172S-AC152-S1/4, Dated 27. August 2007; CSL-07-C172S-AC152-S2/4, Dated 27. August 2007; CSL-07-C172S-AC152-S3/4, Dated 27. August 2007; CSL-07-C172S-AC152-S4/4, Dated 27. August 2007 | | 10.10.2007 |
| 10014989 | CESKA LETECKA SERVISNI A.S. | EASA.A.S.03284 | | | LETECKE ZAVODY L-410-UVP-E13 | Installation of ACAS II system with XPDR (MST-67A) and ELT defined in document CLS-05-L410-AC125, revision 02 dated 04.11.2005 | | CLS-05-L410-AC125, revision 02 dated 04.11.2005 | This STC is valid only for L410-UVP-E13 aircraft type, registration OK-UDS, S/N 892321 | 05.11.2007 |
| 10014990 | CESKA LETECKA SERVISNI A.S. | EASA.A.S.03285 | | | LETECKE ZAVODY L-410-UVP | Installation of NAV/COM/GPS as defined in document CLS-06-6410-AC140, issue 0, revision 2 dated 13.09.2006 | | CLS-06-6410-AC140, issue 0, revision 2 dated 13.09.2006 | 1. This STC is valid only for L410-UVP aircraft type. | 05.11.2007 |
| 10014991 | CESKA LETECKA SERVISNI A.S. | EASA.A.S.03286 | | | LETECKE ZAVODY L 410 UVP-E | Installation of ACAS II (CAS-67) system with one XPDR (MST-67), EGPWS Mk. VI system- KLN-94 GPS receiver, FA 2100 CVR, C 406-1 ELT- KMA-24H-70, LUN1120 and SAE-5-35as defined in document CLS-06-L410-AC138, issue 0, revision 1 dated 14.04.2006 | | CLS-06-L410-AC138, issue 0, revision 1 dated 14.04.2006 | 1. This is valid only for L-410-UVP-E aircraft type. | 05.11.2007 |
| 10015069 | CESKA LETECKA SERVISNI A.S. | EASA.A.S.03395 | | | LETECKE ZAVODY L 200 A LETECKE ZAVODY L 200 D | Installation of new COM, NAV, GPS, XPDR + Encoder, DME, ADF and flight instruments to meet the current technical and operating requirements of IFR category. | | Flight manual supplements in Czech language only: AC159-2501-ELTIME-406AC159-2502 - Audiopanel GMA-347AC159-2503 - NAV/COM/GPS GNS-430AC159-2504 - Radarový odpovídac: GTX-330AC159-2505 - Gyromagnetický kompasový systém KCS-55AAC159-2506 - Meric vzdáleno (...) | | 04.07.2008 |
| 10015193 | CESKA LETECKA SERVISNI A.S. | EASA.A.S.03577 | | | CESSNA 414 CESSNA 414A | Avionics upgrade | | AC172-2501 SUPPLEMENT GARMIN GTX 327 MODE A/C TRANSPONDER (XPDR) AC172-2502 SUPPLEMENT GARMIN GTX 330 MODE STRANSPONDER (XPDR) AC172-2503 SUPPLEMENT ME-406 ELT AC172-2504 SUPPLEMENT Audiopanel GMA-347AC172-2505 SUPPLEMENT (...) | | 16.12.2008 |
| 10029112 | CESKA LETECKA SERVISNI A.S. | | | | L-410 UVP-E-LW, L-410 UVP-E20, L-410 UVP-E9, L-410 UVP-LW, L-410M, L-410 UVP, L-410 UVP-E L-420 | Dual Mode-S Transponder upgrade for EHS capability | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Arndt 1, dated 12/02/2009; Special Conditions: EASA INT POL 23/1 (HIRF) | Flight Manual Supplements: -No. AC186-2501: FMS Unit UNS-1Fw, issued January 2010; -No. AC186-2502: Supplement No.5 - TCAS-94 (SW7.0) a odpovídac: S ModuTDR-94D Rockwell Collins, issued January 2010; -No. AC186-2801: MODE S EHS, issued January 2010 | None | 05.03.2010 |
| 10029701 | CESKA LETECKA SERVISNI A.S. | P-EASA.A.S.03951 | | | SUKHOI SU-31 | Avionics upgrade. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23. | AC190-2501 GARMIN SL 40 Comm Transceiver (COM), December, 2009; AC190-2502 GARMIN GTX 328 mode S transponder (XPDR), December, 2009 | None. | 14.04.2010 |
| 10016012 | CESSNA AIRCRAFT COMPANY | EASA.IM.A.S.02178 | | ST01399W1 | CESSNA 680 CESSNA 650 CITATION III CESSNA 650 CITATION VI CESSNA 650 CITATION VII CESSNA 750 CITATION X | Cessna STC ST01399W1, of 21 August 2006. Installation of a Honeywell DL-950 Data Loader to replace the existing DL-800. | | Approved Model List No. CMR-750-115, of 20 June 2006. Installation Instructions No. CMR-750-111, of 27 Oct. 2005. No AFM Supplements | | 05.02.2007 |
| 10016637 | CESSNA AIRCRAFT COMPANY | EASA.IM.A.S.03077 | | FAA STC SA01519W1 | CESSNA 525 | Installation of an observer seat/3rd audio station/Extended Range Comm. | | 1. According to Cessna Master Document List Doc. No. CF90094 Rev. B, or later Revisions. 2. To be operated in accordance with FAA approved Cessna AFM-S Document No. FM90940, Original Issue, or later approved Revisions. 3. To be maintained in (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 28.11.2008 |
| 10013934 | CESSNA CITATION EUROPEAN SERVICE | EASA.A.S.00334 | REV. 1 | | CESSNA 525 | Installation of a Skywatch HP SKY899A TCAS I coupled to a Honeywell KMD-850 MFD, in accordance with Cessna Citation European Service Center FAM Ref.: 111, Rev NC, Dated 14 August 2007 and FAM ref 82, Rev 01, dated 22 October 2004 or later approved r (...) | | The Airplane Flight Manual Supplements (AFMS) Ref. FM90598 Rev A and CCESC FM 82 Rev A, dated 27th August 2007 or later EASA approved revisions, are required on board for operating the aircraft. The Instructions for Continued Airworthiness Ref CMR-525 (...) | 1. Applicability: Cessna 525 MSN 525-001 to 359 included equipped with Honeywell WX RS-811A and FMS GNS-XLS. | 30.08.2007 |
| 10013961 | CESSNA DUESSELDORF CITATION | EASA.A.S.00645 | | | LEARJET 55 | Installation of Honeywell ACAS II CAS 67A and Mode S XPDR MST 67A | | Engineering Order Report 2004-5, Rev B, dated 3 December 2004; AFMS Report 2004-5-8, Rev A dated 20 October 2004 | 1. FAA TCDS A10CE plus JAA TGL No 8 and JAA TGL No 13 apply. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 17.05.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|----------------|----------------|----------------------|--|---|--|--|--|---------------|
| 10013983 | CESSNA DUESSELDORF CITATION | EASA.A.S.01015 | | | CESSNA 501-551 | Installation of a Honeywell KTA 870 and ART2000 | | Flight Manual Supplement 2003-2-8 Revision A, or later EASA approved revisions. | | 15.06.2005 |
| 10014009 | CESSNA DUESSELDORF CITATION | EASA.A.S.01052 | | | CESSNA 525- 525A | Installation of a Dual MST 67A Mode S Transponder System. | | Airplane Flight Manual Supplement, Report 2004-17-8, Revision A or later EASA approved Revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 13.06.2005 |
| 10014010 | CESSNA DUESSELDORF CITATION | EASA.A.S.01053 | | | 525 CESSNA 525A | Installation of Dual Garmin GTX 330D Mode SAFM Supplement 2004-16-8 Revision A or later EASA approved RevisionMM Supplement 2005-16-10 Revision A or later EASA approved Revision | | | | 17.11.2005 |
| 10014041 | CESSNA DUESSELDORF CITATION | EASA.A.S.01116 | | | PA-42 SERIES | Installation of Artex ELT 406. | | EASA approved Airplane Flight Manual Supplement, Report: 2004-20-8, revision A, or later EASA approved revision. | | 23.06.2005 |
| 10014108 | CESSNA DUESSELDORF CITATION | EASA.A.S.01213 | | | CESSNA 650 | Installation of SATCOM Universal TT-3000 | | Airplane Flight Manual Supplement Doc. No. 2005-18-8 Rev. A or later approved Revision/Maintenance Manual Supplement DocNo. 2005-18-10 Revision A | | 16.11.2005 |
| 10014109 | CESSNA DUESSELDORF CITATION | EASA.A.S.01214 | | | CESSNA 650 | Installation of Airshow Entertainer System | | Engineering Data Report 2005-17, Rev. BAirplane Flight Manual Supplement Report 2005-17-8, Rev. A or later EASA approved revision/Maintenance Manual Supplement Report 2005-17-10, Rev. A | Part 25 of the Federal Aviation Regulations effective February 1, 1965 as amended by Amendments 25-1 through 25-39; TGMJAA leafletNo. 17Weight and Balance Analysis, Report 2005-17-9, Rev. A | 13.03.2006 |
| 10014110 | CESSNA DUESSELDORF CITATION | EASA.A.S.01215 | | | CESSNA 550-560-S550 | Installation of ACAS II CAS 67A and Dual Mode - S Transponder RCZ-833 | | Airplane Flight Manual Supplement Cessna C550, S550, C560 with Honeywell CAS 67A and Dual RCZ 833(), Report No. 2005-1-8 Revision A, dated June 20, 2005 or later approved Revision | | 23.09.2005 |
| 10014144 | CESSNA DUESSELDORF CITATION | EASA.A.S.01262 | | | LEARJET 55 | Installation of Secondary UNS 1F | | Engineering Order 2005-3, Rev. A, or later EASA approved revisions, AFMS 2005-3-8, Rev. A, or later EASA approved revisions | | 23.06.2005 |
| 10014145 | CESSNA DUESSELDORF CITATION | EASA.A.S.01263 | | | LEARJET 55 | Installation of ELT 406 | | Engineering Order 2005-2, Rev. A, or later EASA approved revisions, AFMS 2005-2-8, Rev. A, or later EASA approved revisions | | 29.06.2005 |
| 10014153 | CESSNA DUESSELDORF CITATION | EASA.A.S.01274 | | | GULFSTREAM ASTRA/SPX | Upgrade of Universal UNS-1C to 1C+ | | Engineering Order Report 2005-4, Rev. A dated 22. March 2005 or later EASA approved RevisionsAirplane Flight Manual Supplement Report 2005-4-8, Revision A dated 22. March 2005 or later EASA approved RevisionsMaintenance Manual Supplement Report 2005-4-10 | | 21.09.2005 |
| 10014271 | CESSNA DUESSELDORF CITATION | EASA.A.S.01501 | REV. 1 | | 525, 525A | Installation of Honeywell KMH 880 or KMH-980 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.FAR-23The requirements for environmental protection and the associated certificated noi (...) | Installation Procedure 2003-10-1, Rev. CFlight Manual Supplement 2003-10-8, Rev. DMaintenance Manual Supplement 2003-10-10, Rev. B or later revisions of the above listed documents approved by EASA | none | 27.06.2011 |
| 10014272 | CESSNA DUESSELDORF CITATION | EASA.A.S.01502 | | | CESSNA 501 CESSNA 551 | Installation of Honeywell RDS 84 WX System. | | MM-Supplement 2004-08-10, Rev. A or later approved revision. | | 16.08.2005 |
| 10014288 | CESSNA DUESSELDORF CITATION | EASA.A.S.01529 | | | BEECH 1900 | Installation of Universal Flight Manager System UNS-1L | | -AFM Supplement Beechcraft 1900 and series with Universal UNS-1L; Report 2005-23-08-AMM Supplement for Universal UNS-1L; (et Aviation Document 2005-23-10 | | 25.01.2006 |
| 10014289 | CESSNA DUESSELDORF CITATION | EASA.A.S.01531 | | | BEECH 200 BEECH 200C | Installation of Universal Flight Management System UNS-1L | | - Engineering Order Report 2005-12 Rev. A- AFM-Supplement Beechcraft 200 and Series with Universal UNS-1L Report 2005-12-8 Rev. A or later- Maintenance Manual System Beechcraft 200 and Series with Universal UNS-1L Report 2005-12-10 Rev. A | | 24.10.2006 |
| 10014290 | CESSNA DUESSELDORF CITATION | EASA.A.S.01532 | | | IAI ASTRA SPX | Installation of Universal Cockpit Display System | | Engineering Order, Report 2005-8 dated 23 June 2005, Rev. A or later EASA approved RevisionsAirplane Flight Manual Supplement, Report 2005-8-8 dated 23 June 2005 Rev. A or later EASA approved RevisionsInstruction for Continued Airworthiness, Maintenance Manual Supplement Ref. 2005-8-10 | n/a | 14.12.2005 |
| 10014297 | CESSNA DUESSELDORF CITATION | EASA.A.S.01541 | | | BEECH 200 BEECH 200C BEECH B200 BEECH B200C | Installation of Universal Multifunction Display MFD-640 | | - Engineering Order Reprt 2005-14 Rev. A - AFM- Supplement Beechcraft 200 and Series with Universal Multifunction Display MFD-640 UNS-1L Report 2005-14-8 Rev. A or later approved revision - Maintenance Manual System Beechcraft 200 and Series with (...) | | 31.08.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|----------------|----------------|----------------------|--|---|---------------------|---|--|---------------|
| 10014299 | CESSNA DUESSELDORF CITATION | EASA.A.S.01543 | | | ASTRA 1125 GULFSTREAM 100 | Installation of Collins TCAS 4000 | | Engineering Order, Report 2005-28, Revision B dated 24, April 2006 or later approved Revisions. Airplane Flight Manual Supplement, Rapport 2005-28-8, Revision B dated 18 July 2006 or later approved Revisions | Certification basis of the change: FAR Part 25, additionally JAA Leaflet No. 8 Rev. 2 (ACAS II), JAA Leaflet No. 13 Rev. 1 (Mode S ELS), JAA Leaflet No. 18 (Mode S EHS) | 28.08.2006 |
| 10014300 | CESSNA DUESSELDORF CITATION | EASA.A.S.01544 | | | WESTWIND ASTRA 1125 | Installation of Garmin GNC 300XL | | Engineering Order, Report 2005-29, Revision B dated 24, April 2006 or later approved Revisions. Airplane Flight Manual Supplement, Report 2005-29-8, Revision B dated 24, April 2006 or later approved Revisions. | Certification basis of the change: FAR Part 25, additionally EASA AMC 20-5 (GPS) | 28.08.2006 |
| 10014329 | CESSNA DUESSELDORF CITATION | EASA.A.S.01582 | | | CESSNA 560 (CITATION V) | Installation, Completion of custom Aircraft Interior for BD700-1A10 Global Expressacc.: CDL D954000 Rev. K Validation of imported TCC STC: SA04-113 | | AFM Supplement G-FM25004001 Rev N/C or later TCCA approved Rev. ICA GC91450000-IFCAAMM Supplement G-TM05104001 | Ref. G-FM25004001 Rev. N/C: Limited to SN 9145 | 13.03.2006 |
| 10014345 | CESSNA DUESSELDORF CITATION | EASA.A.S.01608 | | | BEECH 200 BEECH 200C BEECH B200 BEECH B200C | Installation of Blue Sky C-1000 SATCOM | | - Engineering Order Report 2005-15 Rev. A - AFM Supplement for Beechcraft 200 and Series with Blue Sky C-1000 SATCOM Report 2005-15-8 Rev. A or later approved revision. - Maintenance Manual System Beechcraft 200 and Series with Blue Sky C-1000 SATCOM Report 2005-15-10 Rev. A | | 31.08.2006 |
| 10014346 | CESSNA DUESSELDORF CITATION | EASA.A.S.01609 | | | BEECH 200 BEECH 200C BEECH B200 BEECH B200C | Installation of Universal TAWS | | - Engineering Order Report 2005-15 Rev. A - AFM Supplement for Beechcraft 200 and Series with Universal TAWS Report 2005-13-8 Rev. A or later approved revision | | 31.08.2006 |
| 10014351 | CESSNA DUESSELDORF CITATION | EASA.A.S.01616 | | | LEARJET 35 LEARJET 35A | Installation of Flight Data Recorder F1000 | | Engineering Order, 2003-6, Revision B or later EASA approved revisions | | 26.10.2005 |
| 10014366 | CESSNA DUESSELDORF CITATION | EASA.A.S.01636 | | | CESSNA 550 CESSNA 560 CESSNA S550 | Installation of Honeywell CAS 81 ACAS and MST 67A Mode-S XPDR | | Jet Aviation Engineering Order, Report 2005-6, Rev. A Airplane Flight Manual Supplement, Report 2005-6-8, Rev. A or later approved Revision/Maintenance Manual Supplement, Report 2005-6-10 | | 14.11.2005 |
| 10014370 | CESSNA DUESSELDORF CITATION | EASA.A.S.01641 | | | CONSTRUCCIONES C-212 | Installation of CAS 67A ACAS II and dual MST 67A Mode-S XPDR | | Engineering Order, Report 2005-37, Rev. B, 17.01.2006 Airplane Flight Manual Supplement, Report 2005-37-8, Rev. A, Date 22. December 2005 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 23.02.2006 |
| 10014374 | CESSNA DUESSELDORF CITATION | EASA.A.S.01649 | | | BEECH 300 BEECH 300LW BEECH B300 | Installation of Universal UNS-1L Software 802.XX | | AFM Supplement Beechcraft B300 and Series with Universal UNS-1L Report 2005-33-8, Rev. B, or later EASA approved revision | Instructions for Continued Airworthiness, Doc. No. 2005-33-10 Rev. B | 13.07.2006 |
| 10014457 | CESSNA DUESSELDORF CITATION | EASA.A.S.02511 | | | CESSNA 525 CESSNA 525A | Installation of EGPWS MARK VIII | | AFM Supplement Report 2005-9-8, Rev. D, or later EASA approved revision | Maintenance Manual Supplement 2005-9-10-Rev. B | 31.07.2006 |
| 10014458 | CESSNA DUESSELDORF CITATION | EASA.A.S.02515 | | | CESSNA 525 CESSNA 525A | Installation of Skywatch HP | | AFM Supplement Report 2005-10-8, Rev. B, or later EASA approved revision | Maintenance Manual Supplement 2005-10-10-Rev. B | 28.07.2006 |
| 10014541 | CESSNA DUESSELDORF CITATION | EASA.A.S.02628 | | | LEARJET 35 LEARJET 35A | Installation of Universal UNS 1LUK+ | | Engineering Order, Report 2006-3, Revision B dated 10 July 2006 or later approved Revisions. Airplane Flight Manual Supplement, Report 2006-3-8, Revision B dated 10 July, 2006 or later approved Revisions | Certification Basis of the change: FAR Part 25, additionally EASA AMC 20-4 (B-RNAV), EASA AMC 20-5 (GPS), JAA TGL No. 10 Rev. 1 (P-RNAV) | 11.09.2006 |
| 10014553 | CESSNA DUESSELDORF CITATION | EASA.A.S.02651 | | | PIPER PA-46-310P PIPER PA-46-350P | Installation of Garmin GTX 330 or Garmin GTX 330D Transponder | | EASA-approved Airplane Flight Manual Supplement, Report 2004-21-8, Rev. B, dated 01.12.2004, or later EASA-approved edition Maintenance Manual Supplement, Report 2004-21-10, Rev. B | | 27.04.2006 |
| 10014647 | CESSNA DUESSELDORF CITATION | EASA.A.S.02802 | | | LEARJET 35 LEARJET 35A | Installation of Honeywell RDR 2000 and KMD 850. | | Report 2006-16, Rev. D (or later approved revision). AMM Supplement 2006-16-10, rev. A (or later approved revision) | | 25.06.2007 |
| 10014726 | CESSNA DUESSELDORF CITATION | EASA.A.S.02903 | | | CESSNA C850 | Installation of Solid State Flight Data Recorder (SSFDR) F1000 | | Installation in accordance with Master Document Listref MDL-A/C-067 Rev 3 or later revisions. Operation in accordance with Flight Manual Supplementref AFM-060 Rev 3, or later revisions. Instructions for Continued Airworthinessref HB-A/C-711, Rev 2, or later revisions. | | 04.05.2007 |
| 10014741 | CESSNA DUESSELDORF CITATION | EASA.A.S.02924 | | | BEECH 1900 BEECH 1900C WET WING, AIRLINER BEECH 1900D | Installation of an Executive Power System (EPS) with Inverter and Power Outlet. | | # Installation Procedure: 2006-10-1 Rev. B.# AFM-Supplement: 2006-10-8 Rev. C. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 07.09.2007 |
| 10014742 | CESSNA DUESSELDORF CITATION | EASA.A.S.02925 | | | BEECH 1900 BEECH 1900C, AIRLINER BEECH 1900D | Installation of a Satcom System for Passenger Communication. | | # Installation Procedure: 2006-9-1 Rev. B.# AFM-Supplement: 2006-9-8 Rev. A | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 07.09.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|------------------|----------------|----------------------|---|---|---|--|--|---------------|
| 10014922 | CESSNA DUESSELDORF CITATION | EASA.A.S.03169 | | | CESSNA 650 CITATION III,VI,VII | Installation of Honeywell CAS-67A and RCZ-833 Mode S transponders. | | 1. Installed in accordance with Jet Aviation Installation Procedure defined in Engineering Order document ref. 2007-14-1.2. Operated in accordance with Aircraft Flight Manual Supplement (AFMS) ref. 2007-14-8;3. Instructions for continued ai (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 01.02.2008 |
| 10015151 | CESSNA DUESSELDORF CITATION | EASA.A.S.03512 | | | PILATUS PC-12 PILATUS PC-12/45 PILATUS PC-12/47 | Replacement of existing KX 165 COM / NAV, KLN 90B GPS, and GTX 330D Mode S Transponder installation by GNS480 COM / NAV/ GPS and GTX 33D Mode S Transponder installation. | | EASA approved Engineering Data, Doc. No. 2007-23, Revision B, dated 25-Jul-2008, or later approved IssueEASA approved Airplane Flight Manual Supplement, Doc. No. 2007-23-8, Revision A, dated 16-Jan-2008, or later approved IssueMaintenance Manual Sup (...) | Limited to single GNS 480 installation per cockpit only.Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introdu (...) | 17.10.2008 |
| 10015152 | CESSNA DUESSELDORF CITATION | EASA.A.S.03513 | | | PILATUS PC-12 PILATUS PC-12/45 PILATUS PC-12/47 | Installation of AIRCELL SATCOM ST3100 | | 1. EASA approved Engineering Data, Doc. No. 2007-24, Revision C, dated 01-Oct-2008, or later approved Issue.2. EASA approved Airplane Flight Manual Supplement, Doc. No. 2007-24-8, Revision A, dated 11-Jan-2008, or later approved Issue.3. Maintenance (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 14.10.2008 |
| 10015161 | CESSNA DUESSELDORF CITATION | EASA.A.S.03530 | | | CESSNA 525 CESSNA 525A | Upgrade of existing Dual Honeywell MST 67A Mode-S XPDR Installation to EHS Capabilities | | EASA approved Engineering Data, Doc. No. 2007-17, Revision B, dated 01-Jul-2008, or later approved IssueEASA approved Airplane Flight Manual Supplement, Doc. No. 2007-17-8, Revision A, dated 18-Jan-2008, or later approved IssueMaintenance Manual Sup (...) | For variant 525 this change is limited to serial numbers 525 0360 and subsequent | 07.07.2008 |
| 10015171 | CESSNA DUESSELDORF CITATION | EASA.A.S.03545 | | | PILATUS PC-12 PILATUS PC-12/45 PILATUS PC-12/47 | Installation of Honeywell EGPWS MK VI by installing a Honeywell EGPWS MK-VI | | Engineering order, Doc. No. 2007-22, Rev. B or latest approved revision.AFMS, Doc. No. 2007-22-8, Rev. B or latest approved revision | | 13.10.2008 |
| 10015190 | CESSNA DUESSELDORF CITATION | EASA.A.S.03574 | | | CESSNA 525 CESSNA 525A | Installation of GARMIN GPS 400 (w) or 500 (w) | | 1. EASA approved Engineering Data, Doc. No. 2008-2, Revision B, dated 11-Feb-2009, or later approved Issue.2. EASA approved Airplane Flight Manual Supplement, Doc. No. 2008-2 A8, Revision D, dated 03-Mar-2009, or later approved Issue.3. Maintenance Manual Supplement, Doc. No. 2008-2-10, Revision | 1. For approved Software Versions refer to Airplane Flight Manual, Section II - Limitations.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously (...) | 28.04.2009 |
| 10015231 | CESSNA DUESSELDORF CITATION | EASA.A.S.03653 | | | CESSNA CITATION C525 | Installation of AIRCELL Iridium Dual Patch Satcom Antenna. | | 1. EASA approved Engineering Data, Doc. No. 2008-3, Revision B, dated 27-Apr-2009, or later approved Issue- Maintenance Manual Supplement, Doc. No. 2008-3-10, Revision B, dated 27-Apr-2009, or later approved Issue | 1. Antenna Installation only.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon t (...) | 12.05.2009 |
| 10015295 | CESSNA DUESSELDORF CITATION | EASA.A.S.03840 | | | CESSNA 650 | Installation of dual Honeywell GNS-XLS PRNAV FMS | | 1. Engineering Order 2009-2 Revision D from 12 May 2009-AFM Supplement 2009-2-8 Revision B from 12 May 2009-Maintenance Manual Supplement 2009-2-10 Revision A from 3 March 2009-Declaration of Compliance of project 2009-2 from 20 May 2009 or later approved revisions | 1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2) Th (...) | 27.05.2009 |
| 10027115 | CESSNA DUESSELDORF CITATION | P-EASA.A.S.03705 | | | 500, 550, S550, 560 | Installation of a TAWS System Honeywell Mark VIII EGPWS | FAR-25 | Installation Procedure: 2008-11-1, Rev. AFlight Manual Supplement: 2008-11-8, Rev. BMaintenance Manual Supplement: 2008-11-10, Rev. A | Restricted to aircraft which already have installed a suitable multifunction display. | 08.09.2009 |
| 10027116 | CESSNA DUESSELDORF CITATION | P-EASA.A.S.03730 | | | 500, 550, S550, 560 | Installation of an FMS System Universal UNS-1M | FAR-25The Certification Basis for the original product remains applicable to this certificate/ approvalThe certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Installation Procedure: 2008-7-1, Rev. BFlight Manual Supplement: 2008-7-8, Rev. BMaintenance Manual Supplement: 2008-7-10, Rev. Aor later revisions of the above listed documents approved by EASA | None | 08.09.2009 |
| 10027199 | CESSNA DUESSELDORF CITATION | P-EASA.A.S.03743 | | | P.180 AVANTI II | Installation of Garmin GPS-400 / 400W System | The Certification Basis of the original product remains applicable to this „changeThe certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | 1. EASA approved Engineering Order, Doc. No. 2008-13, Revision A, dated 01-Nov-2008, or later approved IssueEASA approved Airplane Flight Manual Supplement, Doc. No. 2008138, Revision A, dated 01-Nov-2008, or later approved Issue-Maintenance Manual Supp (...) | None | 14.09.2009 |
| 10027391 | CESSNA DUESSELDORF CITATION | P-EASA.A.S.03654 | | | 500, 550, 560 | Installation of Honeywell KLN-94-GPS | EASA TCDS IM.A.207The Certification Basis for the original product remains applicable to this certificate/ approvalThe certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | 1. Engineering Data Report 2008-8 Revision E from 25 September 2009- Airplane Flight Manual Supplement 2008-8-8 Revision C from 30 April 2009- Maintenance Manual Supplement 2008-8-10 Revision B from 25 September 2009- Declaration of Compliance to projec (...) | The installation is limited to Cessna models 500, 550 and 560 equipped with Honeywell SPZ 500 Auto Pilot /Flight Director system and RD600 /RD44 Horizontal Situation Indicator system.Limitations according to AFMS 2008-8-8 | 28.09.2009 |
| 10027933 | CESSNA DUESSELDORF CITATION | | | | 525A | Upgrade of existing Dual Garmin GTX 330D Transponder System for EHS | The Certification Basis of the original product remains applicable to this changeThe certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | EASA approved Engineering Order, Doc. No. 2009-6, Revision A, dated 05-Jun-2009, or later approved IssueEASA approved Airplane Flight Manual Supplement, Doc. No. 200968, Revision A, dated 05-Jun-2009, or later approved IssueMaintenance Manual Supplement (...) | None | 17.11.2009 |
| 10028549 | CESSNA DUESSELDORF CITATION | EASA.A.S.03911 | | | BEECH 1900 | | | | | 15.07.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|-------------------|----------------|----------------------|--|--|---|--|--|---------------|
| 10028945 | CESSNA DUESSELDORF CITATION | | | | P.180 AVANTI, AVANTI II | Installation of In-Flight Entertainment System in accordance with Engineering Order, Doc. No. 2009-9 | The Certification Basis of the original product remains applicable to this change. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | EASA approved Engineering Order, Doc. No. 2009-9, Revision B, dated 05-Jan-2010, or later approved Issue-EASA approved Airplane Flight Manual Supplement, Doc. No. 200998, Revision A, dated 05-Oct-2009, or later approved Issue-Maintenance Manual Supplement (...) | None | 22.02.2010 |
| 10035941 | CESSNA DUESSELDORF CITATION | | | | BD-700-1A11 | Cabin Reconfiguration | EASA.IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval (...) | Jet Aviation Declaration of Compliance No. 2011-3 dated 27/07/11. Jet Aviation Engineering Order No. 2011-3 Revision A or later revisions of the above listed documents approved by EASA | Restricted to ASN 9166A prerequisite for this approval is Transport Canada STC No. SA05-60 (FAA STC No ST02140NY) as applicable to ASN 9166. However EASA has not approved this STC for ASN 9166 and if this aircraft is to be entered on an EU register then (...) | 29.07.2011 |
| 10046758 | CESSNA DUESSELDORF CITATION | | | | 650 CITATION III, VI, VII | P-RNAV approval of single or dual UNS-1 Csp | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Cessna Düsseldorf Airplane Flight Manual Supplement, 2012-10-8, Rev A, dated 08 March 2013. Cessna Düsseldorf Master Document List, 2012-10-16, Rev IR, dated 24 September 2013 or later revisions of the above listed documents approved by EASA | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 18.10.2013 |
| 10045283 | CHALLENGER AVIATION PRODUCTS INC. | | | FAA STC SE02352CH | SEE EASA APPROVED MODEL LIST | Direct Oil Filter Replacement - EASA Validation of FAA STC SE02352CH Installation of Challenger cleanable oil filter assembly in accordance with Challenger Aviation Products, Inc. Installation Instructions 1124 revision B, dated October 31, 2009, or later FAA approved revision. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Challenger Aviation Products, Inc. Installation Instructions 1124 revision B, dated October 31, 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision (...)) | 1. Compatibility of this design change with previous modifications must be determined by the installer. 2. Use of Challenger Aviation Products, Inc. Inspection and Cleaning Instructions 1125, Revision C, dated October 31, 2009 or later FAA approved revision (...) | 12.06.2013 |
| 10046408 | CHALLENGER AVIATION PRODUCTS INC. | | REV. 1 | FAA STC SA01669CH | SEE APPROVED MODEL LIST | Installation of Challenger re-usable air filter | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Challenger installation instructions 0402, Revision number as per AML Instructions for Continued Airworthiness 4003, revision D, dated August 31, 2005 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 24.09.2013 |
| 10016805 | CHELTON FLIGHT SYSTEMS | EASA.IM.R.S.01165 | REV. 1 | FAA STC SR02209AK | BELL 206A BELL 206B BELL 206L BELL 206L-1 BELL 206L-3 BELL 206L-4 BELL 407 | Installation of Flight Logic Synthetic Vision Electronic Flight Information System (EFIS-SV). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 150-045709 revision I dated 18 November 2011 - Rotorcraft Flight Manual Supplement or later revisions of the above listed documents approved by EASA. 150-045264 revision AD dated 30 November 2007 or later approved revision - Installation Instructions; 146 (...) | EFIS/AIU integration with the autopilot is approved only when configured with the 902-00028-01 Control Panel. Activation of Fuel Flow functions requires installation of separately approved fuel flow transducers. Installation is not permitted on Instrument Flight Rules (IFR) certified rotorcraft. | 23.11.2011 |
| 10016223 | CHEYENNE TECH | EASA.IM.A.S.02471 | | FAA STC SA01379W1 | PIPER PA-42-1000 PIPER PA-42-720 | Dual Digital Air Data System Installation Qualified for Operation in RVSM Airspace. | | Master Document List MDL8000-34254, Rev. M19-06-2006, Flight Manual Supplement FMS8000-34254, IR27-02-2006, Instructions for Continued Airworthiness ICA8000-34254, Rev. D/22-05-2006. | Additional operational limitations may apply for dual operation due to RVSM Altimeter location. | 06.09.2007 |
| 10015771 | CHIPPEWA AEROSPACE, Inc. | EASA.IM.A.S.01810 | | ST03064AT | BAE JETSTREAM 4101 | Installation of Honeywell MK VI Enhanced Ground Proximity Warning System (EGPWS) | | Chippewa Aerospace Master Drawing List (MDL) J41-3443-0001 Revision IR dated July 29 2003 or later approved revision. Chippewa Aerospace AFM Supplement Document J41-3443-1012 Revision 1 dated May 22 2006 or later approved revision. | | 31.05.2006 |
| 10016183 | CHIPPEWA AEROSPACE, Inc. | EASA.IM.A.S.02412 | | FAA STC ST 03080 AT | | Installation of a Honeywell AAK VI EGPWS. | | # FAA STC ST03080AT. MDL SD3-60-3443-0001 Rev 1 dated 03/28/05.# AFMS SD3-60-3443-1012. Instruction for continued airworthiness SD3-60-3443-0002. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 03.07.2007 |
| 10016184 | CHIPPEWA AEROSPACE, Inc. | EASA.IM.A.S.02413 | | FAA STC ST 02898 AT | BOMBARDIER CL-600-2B19 BOMBARDIER CL-600-2C10 | Installation of a Honeywell MK V EGPWS. (FAA STC ST 02898 AT). | | FAA STC No. ST02898AT, dated 30 July 2004. FAA Statement of Compliance, dated 3 April 2007. Chippewa Aerospace Master Drawing List (MDL) No. 601-3441-0001 Rev. IR, dated 17 May 2004 or later approved revision. Chippewa Aerospace Airplane Flight Manual S (...) | | 07.04.2008 |
| 10016185 | CHIPPEWA AEROSPACE, Inc. | EASA.IM.A.S.02414 | | FAA STC ST 03336 AT | BOMBARDIER CL-600-2B19 | Installation of Protective Covers for K4XD Relay, Traffic Alert and Collision Avoidance Systems (TCAS) Processor and Horizontal Stabilizer Control Unit (HSTCU). | | FAA STC ST03336AT. | N/A | 29.05.2007 |
| 10016186 | CHIPPEWA AEROSPACE, Inc. | EASA.IM.A.S.02415 | | FAA STC ST 03099 AT | BOMBARDIER CL-600-2B19 | Installation of a replacement landing gear wiring harness. | | FAA STC 03099AT | N/A | 29.05.2007 |
| 10016345 | CHIPPEWA AEROSPACE, Inc. | EASA.IM.A.S.02662 | | FAA STC ST2685AT | SAAB 340B SAAB SF340A | Installation of FA2100CVF Cockpit Voice Recorder in Saab 340 series. | | FAA STC ST02685AT | N/A | 22.02.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------|----------------------|----------------|----------------------|---|--|--|--|--|---------------|
| 10016437 | CHIPPEWA AEROSPACE, Inc. | EASA.I.M.A.S.02778 | | FAA STC ST03336AT | BOMBARDIER CL-600-2B19 BOMBARDIER CL-600-2C10 BOMBARDIER CL-600-2D15 BOMBARDIER CL-600-2D24 | Installs of Protective Covers for the 1K4XD, TCAS and HSTU Processor. | | FAA STC No. ST03336AT dated 29 September 2006 FAA Statement of Compliance (undated), Chippewa Aerospace Master Drawing List (MDL) No. 601-2451-01-0001, Rev 1, dated December 07, 2006, or later approved revision. | | 09.04.2008 |
| 10016506 | CHIPPEWA AEROSPACE, Inc. | EASA.I.M.A.S.02870 | | FAA STC ST03573AT | BOMBARDIER CL-600-2B19 BOMBARDIER CL-600-2C10 BOMBARDIER CL-600-2D15 BOMBARDIER CL-600-2D24 | Installs the electrical provisions for Goodrich 8730A2 Class 2 Elect. Fit. Bag. | | FAA Statement of Compliance dated 29 May 2008.FAA STC No. ST03573AT dated 25 April 2008.Chippewa Aerospace MDL No. 601-4620-01-0001 revision IR dated 29 January 2008 or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.This (...) | 24.06.2008 |
| 10026871 | CHIPPEWA AEROSPACE, Inc. | P-EASA.I.M.A.S.03213 | | FAA STC ST03280AT | BAE 125-700 BAE 125-800 BAE 125-800A BAE 125-800B HAWKER 750 HAWKER 800 HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of a Honeywell FileLink Engine Condition Trend Monitoring- Data Downloader (ECTM-DD) | FAR 25 | Installation in accordance with Chippewa Aerospace Master Drawing List (MDL) No H800-7610-01-0001, Rev 11, dated February 26, 2009, or later approved revisions.Instructions for Continued airworthiness in accordance with Chippewa Aerospace Instructions (...) | None. | 19.08.2009 |
| 10026882 | CHIPPEWA AEROSPACE, Inc. | P-EASA.I.M.A.S.03218 | REV. 1 | FAA STC ST03573AT | CL-600-2B19, CL-600-2C10, CL-600-2D15, CL-600-2D24. | Installation of Goodrich Class II EFB System | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRIsH-01 - EWIS.The requirements for environmental protection and the associated certified noise and/ or emissions levels (...) | FAA STC No. ST03573AT dated 05 August 2010 - Chippewa Aerospace Master Data List (MDL) 601-4620-01-0001, Revision 4, dated 26 February 2009.- Chippewa Aerospace Airplane Flight Manual Supplement No.601-4620-0012B Revision IR.- Instructions for Cont (...) | Restricted to Configuration 2 as defined in Chippewa Aerospace Certification Plan No 601-4620-CP Revision D, dated 24 March 2009. | 14.01.2013 |
| 10031668 | CHIPPEWA AEROSPACE, Inc. | | REV. 1 | FAA STC SA03859AT | LIBERTY XL-2 | Garmin GTX 330 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged (...) | Chippewa Aerospace Master Data List (MDL) XL2-3454-01-001, Rev. IR, dated 02/10/2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None | 05.11.2010 |
| 10032241 | CHIPPEWA AEROSPACE, Inc. | | | FAA STC SA03860AT | LIBERTY XL-2 | Honeywell KN-62A DME | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged (...) | Honeywell KN-62A DME Instructions for Continued Airworthiness XL2-3453-01-002AFMS XL2-3453-01-1012 Rev IR dated July 06 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | none | 05.11.2010 |
| 10042101 | CHIPPEWA AEROSPACE, Inc. | | | FAA STC ST02604AT | DHC-8-102,-103,-106 DHC-8-201,-202 DHC-8-301,-311,-314,-315 | Installation of a Honeywell MK VIII Enhanced Ground Proximity Warning System (EGPWS) | FAR 25 as per EASA.I.M.A.191, Issue 6 (incl. EWIS)The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/ or emissions levels (...) | Installation: Must be in accordance with Chippewa Aerospace Inc. Master Drawing List B8-3443-1001-(-,-,-,-) Rev. IR, dated 28. October 2002. Operation: Must be in accordance with approved Chippewa Aerospace Inc. FlightManu (...) | The aircraft must be equipped with the factory installed Honeywell Ground Proximity Warning System (GPWS). | 08.11.2012 |
| 10042660 | CHIPPEWA AEROSPACE, Inc. | | | FAA STC ST02046AT | DHC-8-102, DHC-8-103, DHC-8-106, DHC-8-201, DHC-8-202, DHC-8-301, DHC-8-311, DHC-8-314, DHC-8-315. | Installation of a ground service bus FAA STC ST02046AT | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation: Must be in accordance with Aviation Technologies Group Master Drawing List DHC8-0001, dated 20.05.1999 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 12.12.2012 |
| 10014984 | CHRISTEN-AIRTECH AG | EASA.A.S.03276 | | | CESSNA P210N | Avionics Retrofit: King KMA-24; Sigtronics SPA-400; Dual King KY-196; Dual King KN-53; King KI-525A; trimble TNL-2000; King KT-76A; ARGUS-5000; BF Goodrich WX-1000In: Garmin GMA-347; Dual Garmin GNS-430; Garmin GTX-330; Garmin GMX-200; L-3 WX-500. (...) | | Master Document List No.: 207.1603-EOMDL-01-00AFM Supplement GNS-430 No. 1; Doc. No.: 207.1603-FS-01-00AFM Supplement GNS-430 No. 2; Doc.No: 207.1603-FS-02-00 | Reg: HB-CZK S/N: P21-00860 | 24.01.2008 |
| 10014987 | CHRISTEN-AIRTECH AG | EASA.A.S.03280 | | | CESSNA P210N | Avionics Upgrade by removing: 1. GPS-155 GPS2. KX-165 COM/NAV 13. KX-155 COM/NAV 24. WX-8 Stormscope5. KMA-24 Audio Panel and installing: 1. GNS-530 COM/NAV/GPS 12. GNS-430 COM/NAV/GPS 23. WX-500 Stormscope4. GMA-347 Audio Panel | | 1. Declaration of Compliance, Doc. No. 207.1599-DC-01-002. Master Document List, Doc. No. 207.1599, Rev. 0 or latest approved Revision 3. AFMS, Doc. No. 207.1599-FS-01-01, Rev. 1 or latest approved Revision 4. AFMS, Doc. No. 207.1599-FS-02-00, Rev. 0 or latest approved Revision | 1. Reg: HB-CHJ S/N: P210-0522. Valid for this S/N Only 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will intro (...) | 06.11.2007 |
| 10015066 | CHRISTEN-AIRTECH AG | EASA.A.S.03392 | | | BEECH C90 KING AIR | GNS 530A, GNS 430A, GAD 42 installation: Removed equipment: -2 KTR 905 COM 1 & COM 2 - 2 KNR 630 NAV 1 & NAV 2 - 1 KLN 90B GPS with AK-850 GPS Annunciator - 1 KDM 705A DME with KDI 570 DME Indicator New installed equipment: - 1 GNS 530AW COM/NAV/GPS 1 with (...) | | Master Document List, Doc. No. 207.1698 Revision 0 or latest approved Revision: GNS 530AW-AFMS, Doc. No. 207.1698-FS-01-00 Rev. 0 or latest approved Revision- GNS 430AW-AFMS, Doc.No. 207.1698-FS-02-00, Rev. 0 or latest approved Revision | F-GBLU, SN: LJ-822. | 10.09.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|-------------------|----------------|----------------------|--|---|---|--|---|---------------|
| 10015129 | CHRISTEN-AIRTECH AG | EASA.A.S.03480 | | | CESSNA P210N | Avionics upgradeBy removing:-NSD 360 HISSA-832B Slaving BoxACK A-30 mod 8 Altitude EncoderGMA-340 Audio PanelAnd installingSN-3500 E-HISSAC 7-35 Airdada ComputerKG-102A Remote GyroGMA-347 Audio Panel | | Master Document List, Doc. No. 208.1730-AFMS Doc. No. 208.1730-FS-01-00 revision 0 or latest approved revision.AFMS, Doc. No. 208.1730-FS-02-00 revision 0 or latest approved revision. | Reg.: HB-CZB S/N: P-210-0780 | 05.03.2008 |
| 10015262 | CHRISTEN-AIRTECH AG | EASA.A.S.03720 | | | CESSNA 210 | Avionics UpgradeBy removing the existing:-COM/NAV 1 + 2- DME- Mode A/C Transponder- Audio PanelAnd install the new Avionics Systems:- GNS 530W COM/NAV/GPS 1- GNS 430W COM/NAV/GPS2 with G1106A GPS/NAV Indicator- KN 62A DME- GTX 330 Mode S Transpond (...) | | -Master Document List, Doc. No. 208.2027-Rev. 0 or latest approved Revision- AFMS GNS 530W COM/NAV/GPS 1, Doc. No. 208.2027-FS-01-00, Rev. 0 or latest approved Revision- AFMS GNS430W COM/NAV/GPS 2, Doc. NO. 208.2027-FS-02-00, Rev. 0 or latest appro (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 11.12.2008 |
| 10031914 | CHRISTEN-AIRTECH AG | | | | TBM 700 A | Avionics UpgradeOut: Dual COM-NAV Bendix-King KX-165A, GPS Apollo MNS2101, MFD ApolloMX20, Encoder Shadin 8800-Tin: COM-NAV-GPS Garmin GNS-530W & 430W, MFD Garmin GMX-200, EncoderACKA-30 Mod. 8 serial | CS 23The Certification Basis for the original product remains applicable to this certificate/ approvalThe certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Document List CAT-207.1698-EOMDL-01 Rev. 0 of 15.06.2010Instructions for Continued Airworthiness CAT-210.2483-CA-01 Rev. 0 of 15.09.2010AFM Supplement Doc. No.: 210.2483-FS-01(530W)-00 & -02(430W)-00or later revisions of the above listed documents approved by EASA | Reg.: D-FIRE, S/N: 137 | 21.09.2010 |
| 10036867 | CHRISTEN-AIRTECH AG | | | | 414A | Inst.Traffic alerting system/StormscopeRemoved:- NilInstalled:- Garmin GTS800 Traffic Alerting System- L-3 WX500 Stormscope System | CS 23, Amdt: 2The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approvalSCs or ESFs: For this modification, voluntary compliance to CS23 Amdt. 2 (...) | Master Document List: Doc. Nr. CAT-211.2949-EOMDL-01, Rev. 0Instructions for continued airworthiness:- For the Garmin GTS 800, Doc. Nr. CAT-211.2949-CA-01-For the L-3 WX500, Doc Nr. CAT-211.2949-CA-02 Rev 0 | S/N: 414A-0233,,D-IMUB | 13.10.2011 |
| 10015968 | CIRRUS DESIGN CORPORATION | EASA.IM.A.S.02119 | | SA01355W1-D | CIRRUS SR20 | Validation of FAA STC SA01355W1-D # #Installation of Bendix/King KR87 ADF and/ or KN 62A DME# | | Installation in accordance with Master Drawing List 159-08310-0001, Rev. C, or later FAA approved Revision | 1. The installer must determine, if this design is compatible with previously approved modifications. If the holder agrees to permit another person to use this certification or alter a product, the holder must give the other person written evidenc (...) | 15.05.2007 |
| 10016118 | CITATION TECH | EASA.IM.A.S.02310 | | FAA STC ST01395W1 | CESSNA 500 CESSNA 501 CESSNA 550 CESSNA 551 REIMS AVIATION CESSNA F 177 RG | Installation of Dual Honeywell AM-250 Digital Altimeter andSkylight Avionics Air Data Interface Adapter ADC-602 AIU and associatedequipment to obtain RVSM approval. | | Installation in accordance with Master Document List MDL2703-34254, dated 19 June, 2006 or later approved installation.Compliance Checklist CR2703-34254.AFM Supplement FMS2703-34253-4, Rev. IR, dated 06 June 2007. | | 04.07.2007 |
| 10016006 | CLIPPER AVIATION, Inc. | EASA.IM.A.S.02170 | | FAA STC ST02318NY | BOEING 747-400 FREIGHTER | Upper Deck seating arrangement for B747-400BCF as defined in Weber Aircraft installation drawing No. 860014 and ClipperAviation CLI-2006-003 Rev. A, Master Data List B747-400BCF | | CLI-2006-007 Airplane Flight Manual Supplement for Boeing 747-400 Special Freighter | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and #Associated Technical Documentation#. Compatibility with other aircraft/engine configurations shall be determined by theinstaller. | 04.04.2007 |
| 10036123 | CLIPPER AVIATION, Inc. | | | FAA STC ST02318NY | B747-400 | Installation of seating configuration on Upper Deck of B747-400 Boeingconverted Special Freighter | The Certification Basis for the original product remains applicable to this certificate/ approval. | Master Data List CLI-2006-003, Revision D, dated May 23, 2011Airplane Flight Manual Supplement CLI-2007-015 Rev. I.Ror later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 12.08.2011 |
| 10015822 | COLUMBIA AVIONICS & AIRCRAFT | EASA.IM.A.S.01894 | | ST01128 | BEECH 200 BEECH 200C BEECH 200CT BEECH 200T BEECH 300 BEECH 300LW BEECH B200 BEECH B200CT BEECH B200T BEECH B300 BEECH B300C | Installation of Honeywell MK-VIII Enhanced Ground ProximityWarning System (with or without optional EFIS-85B/14 Display). | | Installed in accordance with Columbia Avionics Inc Master Document List No 30141005, Revision 2, dated December 30, 2002, or later approved revisions.Operation in accordance with FAA approved Flight Manual Supplement 30141020, dated January 8, 2003 or later approved revision. | | 28.09.2006 |
| 10015860 | COLUMBIA AVIONICS & AIRCRAFT | EASA.IM.A.S.01973 | | FAA STC SA01156W1 | CESSNA 501 CESSNA 551 | Installation of dual RVSM compliant IS&S or J2 (Kollsman) air data system. | | According to Columbia Avionics Master Document List Doc. No. 30141024, Rev. 4 or later FAA approved RevisionsTo be operated in accordance with either of- FAA approved Columbia Avionics AFM-S Document No. 30141039, original issue, or later FAA app (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 12.02.2009 |
| 10016271 | COLUMBIA AVIONICS & AIRCRAFT | EASA.IM.A.S.02555 | | FAA STC ST01149W1 | CESSNA 500 CESSNA 550 CESSNA 560 CESSNA S550 | Installation of IS&S or J2 (Kollsman) Air Data Systemwith Reduced Vertical Separation Minimum (RVSM) | | Master Doc. List 30140946 of May 10, 2005 and either one of30140961, IS&S Air Data Flight Manual Supplement -500/S550 of03/12/200330140986,IS&S Air Data Flight Manual Supplement #550/560 of 03/12/200330141197,IS&S Air Data Flight Manual Sup. for KFC- (...) | | 27.08.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|---------------------|----------------|-----------------------|---|---|---|--|---|---------------|
| 10016272 | COLUMBIA AVIONICS & AIRCRAFT | EASA.IM.A.S.02557 | | FAA STC ST01252W1 | CESSNA 560 CESSNA S550 | Vertical Separation Minimum (RVSM) group airworthiness approval of Cessna Modell S550 and 560 Citation Aircraft | | Master Doc. List 30141243, Rev. 3 of June 4, 2003 Instructions for initial and continued airworthiness, Doc.30141246, Rev. A of April 23, 2004, or later FAA approved revisions to these two documents. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 27.08.2007 |
| 10016320 | COLUMBIA AVIONICS & AIRCRAFT | EASA.IM.A.S.02629 | REV. 1 | FAA STC ST01162W1 | CESSNA 500 CESSNA 501 CESSNA 550 CESSNA 551 | Vertical Separation Minimum (RVSM) group airworthiness approval of Cessna Models 500, 501, 550 and 551 Citation Aircraft | | FAA STC ST01162W1, Master Document List 30140986, revision 11, Airplane Flight Manual Supplements 30141699, revision N/C (Model 500, 501, 551), 30141700, revision 1 (Model 550), 30141701, revision 1 (Model 550), 30141702, revision 1 (Model 500), (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 19.03.2009 |
| 10016321 | COLUMBIA AVIONICS & AIRCRAFT | EASA.IM.A.S.02630 | | FAA STC ST01084W1 | | Installation of Collins TCAS-4000 TCAS II System (with or without optional Avidyne Flight Max 800 MFD. | | Installation in accordance with Columbia Avionics, Inc. Master Document List No. 30140855, Revision 2, dated May 31, 2002, or later approved revisions. Operation in accordance with FAA approved Manual Supplement 30140871, dated June 7, 2002 or later Approved Revisions. | | 03.06.2008 |
| 10015895 | COMMUTER AIR TECHNOLOGY Inc. | EASA.IM.A.S.02015 | REV. 1 | FAA STC SA00184LA | BEECH 200 BEECH 200C BEECH 200CT BEECH A200 BEECH A200C BEECH A200CT BEECH B200 BEECH B200C BEECH B200CT BEECH B200T | Installation of Alternate Main Landing Gear Tires, Michelin P/N 026-335-0. Revision 1 issued to add models B200GT and B200CGT. | | Installation in accordance with Commuter Air Technology Top Drawing LE10000 Revision NC, dated May 9, 1996 or later approved revisions. Operation in accordance with Commuter Air Technology Approved Flight Manual Supplement LE-10001, dated May 14, 1996 or later approved revisions. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 10.02.2009 |
| 10016204 | COMMUTER AIR TECHNOLOGY Inc. | EASA.IM.A.S.02446 | | FAA STC SA10370SC | RAYTHEON A36 | Installation of exhaust gas extractors on Pratt & Whitney Canada PT6A-21, -27, -28, -29, -34, -34B, -135, or 135A turboprop engines. | | Installation in accordance with Commuter Air Technology Drawing List No. DL02001, Rev. 8, dated October 8, 2004, or later approved revision. | Note that the airspeed and shaft horsepower limits listed in the original Airplane Flight Manuals are still applicable (see TC DataSheet No. 3A20). Compatibility of this design change with previously approved modifications must be determined by the (...) | 21.05.2007 |
| 10046102 | COMMUTER AIR TECHNOLOGY Inc. | EASA.IM.A.S.03108 | | FAA STC SA2735CE | BEECH 100 AND A100 | Installation of External Lower Fuselage Cargo Compartment | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with DeVore Drawing List Number 100DL, FAA approved dated July 2, 1991. Operate in accordance with FAA approved Flight Manual Supplement, dated July 29, 1991, for Beech Models 100 and A100 with Commuter Air Technology, inc. 60 (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 14.08.2013 |
| 10029892 | COMPAGNIE FRANCE AVIATION | P-EASA.A.S.03869 | | | GARDAN GY80-160 | Installation of Aspen Avionics EFD1000 PRO 1 Garmin GMA340, 1 Garmin GNS 430 and 1 Kannad 406MHZ AF Compact in the GARDAN GY-80 series. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AFM temporaire N1AFM temporaire N2 Limitation VFR de jour or later revisions of the above listed documents approved by EASA | None. | 30.04.2010 |
| 10033572 | COMPAGNIE FRANCE AVIATION | P-EASA.A.S.03031 | | | R2160 | FA 07/2007 Edition 1 - Installation malonnier sur R2160 | none | AFM-07 -2007 -003 dated 11 Sept 2008 AMM-07 -2007 -001 dated 13 Feb 2009 | Limited to S/N 154 | 28.07.2009 |
| 10015540 | COMTEK ADVANCED STRUCTURES Ltd. | EASA.IM.A.S.01424 | | | DORNIER D328-100 | Replacement of trolley storage unit panels | | Comtek Aircraft Maintenance Manual Supplement COMAMM-030, Rev B, TCCA accepted 30 September 2003 or later approved version for installation/removal. Aircraft Maintenance Manual Supplement COMAMM-031 Rev A, TCCA accepted 30 September 2003 or later approved version for maintenance. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.07.2005 |
| 10034405 | COMTRAN INTERNATIONAL, INC. | P-EASA.IM.A.S.02963 | | FAA STC SA00664DE | DORNIER 328-300 | Cabin Interior Re-Configuration Dornier 328-300 | | -STC FAA ST00664DE Comtran International Documents - Master Document List 001-MDL001, Rev. C dated March 20, 2008 - Master Document List 001-MDL002, Rev. A dated March 28, 2008 - Flight Manual Supplement 001-FM001, Rev. IR FAA approved May, 222008 - Instr (...) | None | 29.07.2009 |
| 10015738 | CONCORDE BATTERY CORPORATION | EASA.IM.A.S.01762 | | FAA STC ST00640LA | HAWKER 1000 | Installation of Concorde Battery Model No RG-380E/60K sealed lead acid batteries | | In accordance with Master Drawing List No 20, Revision N/C dated March 10, 1998 or later approved revisions | -- | 06.04.2006 |
| 10016315 | CONCORDE BATTERY CORPORATION | EASA.IM.A.S.02620 | | FAA STC NO. ST01173WI | LEARJET 45 | Installation of Concorde RG-443 Lead Acid Battery. | | None | | 05.02.2008 |
| 10038867 | CONCORDE BATTERY CORPORATION | | | FAA STC 03930AT | ATR 42-200 /300 /320 /500 ATR 72-101 / -102 ATR 72-201 / -202 / -211 ATR 72-212 / -212A | Installation of a Lead Acid Battery This STC installs a Lead Acid battery in replacement of the existing Nickel-Cadmium battery. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: -TCDS EASA.A.084, -CS 25-1729 (Amdt 5) The requirements for environmental protection (...) | Master Drawing List (MDL), ref : 5-0372, Revision D, dated 16 September 2011; Instruction for Continued Airworthiness, ref : 5-0445, Revision A, dated 13 September 2011; Aircraft Flight Manual Supplement, ref : 5-0374, dated 20 June 2011; or later revisi (...) | None. | 20.03.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|--------------------|----------------|------------------------------|---|---|---|---|--|---------------|
| 10040738 | CONCORDE BATTERY CORPORATION | | | FAA STC NO. SA01633W1 | EMB 500 | Installation of Concorde RG-390E VRSLA Batteries in the Embraer 500 in accordance with Concorde Battery Corporation Master Drawing List, DWG No. 5-0359, Revision NC, dated August 25, 2010 or later FAA approved revisions. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Concorde Battery Corporation Master Drawing List, DWG No. 5-0359, Revision NC, dated August 25, 2010; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None. | 24.07.2012 |
| 10040781 | CONCORDE BATTERY CORPORATION | | | FAA STC ST01643W1 | DHC-8-102, -103, -106 DHC-8-201, -202 DHC-8-301, -311, -314, -315 | Installation of Concorde RG-380E/44 (42AH) or RG-380E/60L (48AH) Series Valve Regulated Sealed Lead-Acid (VRSLA) batteries as replacement of OEM batteries. FAA STC ST01643W1 | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Must be in accordance with Concorde Battery Corporation Master Data List (MDL) Drawing N° 5-0284-1, 5-0284-3 Rev. A, Operation. Must be in accordance with Airplane Flight Manual Supplement (AFMS) Drawing N° 5-0285, Rev. NC, Maintenance: Must (...) | none | 26.07.2012 |
| 10042099 | CONCORDE BATTERY CORPORATION | | | FAA STC SA01118W1 | SEE EASA APPROVED MODEL LIST | Installation of the Concorde FLA or VRSLA RG series batteries in Cessna Single Engine Airplanes listed on the attached Approved Model List (AML) in accordance with Concorde Battery Corporation, Master Drawing List Dwg. No. 5-0148, Revision A, dated August (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Concorde Battery Corporation, Master Drawing List Dwg. No. 5-0148, Revision A, dated August 16, 2002; Concorde Battery Corporation, Instructions for Continuing Airworthiness for the RG series batteries Doc. No. 5-0171, Revision M, dated March 15, 2012; Co (...) | None. | 08.11.2012 |
| 10043338 | CONCORDE BATTERY CORPORATION | | | FAA STC SA03971AT | DHC-6 SERIES 1, 100, 200, 300 | Replace NiCd or Lead Acid battery with Concorde Lead Acid battery and associated hardware in accordance with Master Drawing List 5-0398 Revision E, dated May 4, 2012 or later FAA approved drawing revisions. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Instructions for Continued Airworthiness (ICA), Concorde Battery Corporation No. 5-0446, Rev. NC, dated September 29, 2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent r (...) | none | 22.01.2013 |
| 10044684 | CONCORDE BATTERY CORPORATION | | | FAA STC ST00732DE | 328-100 | Replacement of original equipment nickel cadmium batteries with Concorde Model RG-380E/44LS Lead Acid batteries. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Concorde Battery Corporation Master Drawing List ref 5-0353, Revision B, dated September 15, 2011. Operation in accordance with Concorde Battery Corporation FAA Approved Flight Manual Supplement ref 5-0354, Revision B, d (...) | This modification has been reviewed and it has been determined that it does not necessitate a revision to the EWIS ICA that was required to be developed by CRI H-1. | 29.04.2013 |
| 10044950 | CONCORDE BATTERY CORPORATION | | REV. 1 | FAA STC SA01260W1 | SEE ATTACHED FAA AML | Installation of the Concorde RG-380E/K and RG-380E/L series VRSLA batteries in the Beechcraft King Air Model Airplanes. Replacement of the existing aircraft main batteries with a Concorde RG-380E/K or RG-380E/L series Lead Acid Battery. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation of the Concorde RG-380E/K and RG-380E/L series VRSLA batteries in the Beechcraft King Air Model Airplanes (see AML) in accordance with Concorde Battery Corporation, Master Drawing List, Drawing No. 5-0177, Rev. N/C, dated May 1, 2003; Instru (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 24.05.2013 |
| 10045252 | CONCORDE BATTERY CORPORATION | | REV. 1 | FAA STC SR03895AT | MD900 | Installation of an Concorde replacement lead acid battery for MD900 Series | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. FAR Part 27 (as defined in FAA TCDS H19NM); 27.1301(a), (b), (c) & (d), 1309(a) & (b), 1353(a), (b), (c), (d) & (e), 1529. The requirements for enviro (...) | RFM Supplement: 5-0330 23 Nov 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) # and/or the Technical Implementation Procedures of EU/USA Bilateral (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 21.06.2013 |
| 10014097 | CONTOUR AEROSPACE LIMITED | EASA.A.S.01195 | REV 1 | | BOEING 767-300 SERIES | Introduction G3 Wardrobe to British Airways aircraft as defined in the modification definition document and Declaration of Design and Performance (DDP) | | -Heath Teona Modification Document C936 and associated DDP dated 24/3/2005 or later approved revisions. | As defined in the DDP. | 20.05.2005 |
| 10016274 | CORPORATION LOGIC AIR Inc. | EASA.I.M.A.S.02560 | REV 1 | TC STC C-LSA07-205/D ISSUE 2 | BOEING 767-300 | Installation of In-Flight Entertainment System. | | 1. Aero Consulting Services Modification Data Summary F371000, rev A.2. Instruction for Continued Airworthiness: F371071, rev NC for S/N 27957 and 26263; F371072, rev NC for S/N 27135 and 37136. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 10.04.2008 |
| 10016283 | CORPORATION LOGIC AIR Inc. | EASA.I.M.A.S.02578 | | TCCA SA07-210/D ISSUE 1 | BOEING 767-300 | Installation of 270 Pax Interior Configuration. | | Aero Consulting Services Modification Data Summary F276000, rev N. Instruction for Continued Airworthiness, F276071, rev NC. | | 02.11.2007 |
| 10016284 | CORPORATION LOGIC AIR Inc. | EASA.I.M.A.S.02579 | REV 1 | TCCA STC SA07-12 ISSUE 2 | BOEING 767 | Tire pressure Indication System De-activation, Boeing 767-300. | | 1. Aero Consulting Services Modification Data Summary E909000, rev. A.2. Instructions for Continued Airworthiness contained in Maintenance Manual Supplement E909071, rev. C. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 07.08.2008 |
| 10016285 | CORPORATION LOGIC AIR Inc. | EASA.I.M.A.S.02580 | | TCCA STC C-LSA07-091/D | BOEING 767-300 | Installation of Mode S ATC Transponder with Elementary and Enhanced Surveillance Modes. | | Aero Consulting Services Modification Data Summary F278000, rev N/C. Flight Manual Supplement F278092, rev N/C. Instructions for Continued Airworthiness Supplement E278071, rev N/C | | 02.11.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|-----------------------------|--|---|--|---|--|---------------|
| 10016290 | CORPORATION LOGIC AIR Inc. | EASA.IM.A.S.02585 | | C-LSA03-498/D | BOEING 767-300 | Changes to Altitude Callout Configuration, Honeywell Mark VEGPWS. | | Aero Consulting Services Modification Data Summary D799000, rev B. Instructions for Continued Airworthiness Supplement D799071, rev N/C. | | 02.11.2007 |
| 10043626 | CORPORATION LOGIC AIR Inc. | EASA.IM.A.120 | | TCCA STC SA111-92, ISSUE 03 | 737-300/737-400/737-500 | Installation of Dual CMC SBAS Landing System with CMA-5024 GLSSU sensors and Dual CMC CMA-5025 control panels. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ACS-NAI/EMTEQ Documentation:- Modification Data Summary H291000 Rev B- Compliance checklist H291065 CCL Rev B- Aircraft Flight Manual Supplement H291091 Rev N/C- AMC 20-28 Compliance Report H291021A_AMC20-28 Rev B/LOGICAIR Maintenance Manual Supplement (...) | The CMC Electronics CMA-5024 SBASSU and CMA-5025 SBAS CP Operator's Manual No. 930-500105-000, Revision 2 dated September 16, 2009, or later revision, must be available to the flight crew whenever navigation is predicated on the use of the SBAS landing (...) | 13.02.2013 |
| 10033973 | CRANE AEROSPACE & ELECTRONICS | | | FAA STC ST02316LA | LEARJET 60 | Installation of Tire Pressure Sensor FAA STC ST02316LA (Date of issuance: September 28, 2010) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Refer to Master Data List 83-216-01, rev. K for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revision of this decision). | None | 23.02.2011 |
| 10014384 | CRANFIELD AEROSPACE LTD | EASA.A.S.01669 | | | | Installation of a Honeywell Global GNS-XLS | | AFM Supplement - Cranfield Supplement No 2 Issue 11 type Design Definition Addendum CA/DD/E2184 Issue 4DF/ Jetstream 41/1232 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 20.02.2006 |
| 10027822 | CRANFIELD AEROSPACE LTD | P-EASA.A.S.03734 | | | BAE JETSTREAM 3100 | Installation of Satcom System | BCAR D The Certification Basis for the original product remains applicable to this certificate/ approval | Cranfield Modification No Caem/Jetstream/1286 Type Design Definition Addendum CA/DD/DR0054 Issue A or later revisions of the above listed documents approved by EASA | None | 09.11.2009 |
| 10037359 | CRANFIELD AEROSPACE LTD | | | | CESSNA 421C | Installation of Initial Role Change Equipment in a Civil Contingency Aircraft - Cessna 421C as defined in Cranfield Aerospace document CAEM/Cessna/1327. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 23, Amdt. 0-23.21, 23.147, 23.171, 23.179, 23.233, 23.251, 23.303, 23.305, 23.307, (...) | Type Design Definition CA/DD/DR0109AFMS: Supplement No. 2 to POH D1546-5-13 Instructions of Continued Airworthiness CA/DD/ML1327 or later revisions of the above listed documents approved by EASA. | S/N 421C-0218. | 22.11.2011 |
| 10037360 | CRANFIELD AEROSPACE LTD | | | | CESSNA 421C | System and Structural Provisions for a Civil Contingency Aircraft - Cessna 421C as defined in Cranfield Aerospace document CAEM/Cessna/1326. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 23, Amdt. 0-23.21, 23.147, 23.171, 23.179, 23.233, 23.251, 23.303, 23.305, 23.307, (...) | Type Design Definition CA/DD/DR0108AFMS: Supplement No. 1 to POH D1546-5-13 Instructions of Continued Airworthiness CA/DD/ML1326 or later revisions of the above listed documents approved by EASA. | S/N 421C-0218. | 22.11.2011 |
| 10044567 | CRANFIELD AEROSPACE LTD | | | | 421C | Structural Provisions of NELS Antenna Installation in accordance with CAEM/Cessna/1360 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: FAR 23 Amdt. 23-34; 23.21(a); 23.239(a); (b); 23.45(b); 23.141, 23.173; 23.251; 23.303; 23.305(a); (b); 23.307(a); 23.333(a) (...) | Installation Instructions CA/DD/SB1360 Pilot Operating Handbook Supplement No. 8, issue 1 Instructions for Continued Airworthiness CA/DD/SB1360 or later revisions of the above listed documents approved by EASA | Limited to Serial Number 421C-0699 and 421C-0603 The installer must determine whether this design change is compatible with previously approved modifications. If the holder agrees to permit another person to use this certificate to alter the product, the (...) | 19.04.2013 |
| 10014485 | CTT SYSTEMS AB | EASA.A.S.02547 | REV. 2 | CAA NL STC SA0009NL | BOEING 767-300 | Replacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via CAA NL STC SA 0009 NL due to the potential for combustion of contaminants in the dryer unit. A new (...) | | SB 3000-002-01-AD Initial / SB 3000-000-21-07-AD Initial EASA AD 2006-0140-E or later approved revisions Note: This revision reflects an amendment to the installation instructions and supersedes STC EASA.A.S.02547/Revision 1. | Instruction for the installation: Installation instructions I1767-2 and I1767B-2, latest issue. KLM MO 21-0476 and Martinair MO 76730MP-003. Instructions for operation: No specific Airplane Flight Manual Supplement is needed. Prior to installation of this mo (...) | 14.04.2009 |
| 10015267 | CTT SYSTEMS AB | EASA.A.S.03738 | | CAA NL STC SA 0003 NL | BOEING MD 11 | Replacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via CAA NL STC SA 0003 NL due to the potential for combustion of contaminants in the dryer unit. A new (...) | | SB 3000-002-21-01-AD Initial EASA AD 2006-0140-E or later approved revisions | Instruction for the installation: Installation instruction IIMD11, latest revision, Swissair Engineering Order 510440. Instructions for operation: Airplane Flight Manual Supplement, AFM-98-02, Rev. 0, dated 3-11-98 or latest revision. Instructions for co (...) | 15.12.2008 |
| 10016071 | CUB CRAFTERS INC. | EASA.IM.A.S.02252 | REV. 1 | FAA STC SA00522SE | PA-18-125, PA-18A, PA-18-135, PA-18A-135, PA-18-150, PA-18A-150 | Installation of Lycoming O-320-A2B, O-320-A2C, O-320-A2D, O-320-B2B, O-320-B2C, O-320-D2A, O-320-D2B and O-320-D3G engines with Sensenich M74DM-52 through -56 or 74DM6-0-52 through -56 propellers and associated hardware, i.a with FAA STC SA00522SE. | | FAA approved AFM Supplement 103000-FMS150, Rev. A or later FAA approved revision. FAA approved AFM Supplement 103000-FMS160, Rev. A or later FAA approved revision. | The compatibility of this modification with other approved modifications has to be ensured by the company installing the modification. Approved MTOM up to 794 kg. | 11.12.2006 |
| 10042813 | CUSTOM FIBERGLASS MOLDING INC. | | | FAA STC SA02391CH | BEE DEE M-4, M-4 M-4-210, M-4-210C M-4-220, M-4-220C M-4-220S, M-4-180V M-4C, M-4S, M-4T M-5-235C, M-5-180C, M-6-235 M-7-235, M-7-235B MX-7-160, MX-7-180 MX-7-180A, MX-7-235 MX-7-180B, MX-7-180C | Installation of Custom Fiberglass Molding Skis, Model 3000, according to Master Drawing List CFMI 3000-9, Rev D dated 04th April 2011 or later FAA approved revision. Installation of Custom Fiberglass Molding Skis, Model 3000-01(A/B), according to Master D (...) | Design: Non-significant change The Certification Basis of the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation of Custom Fiberglass Molding Skis, Model 3000, according to Master Drawing List CFMI 3000-9, Rev D dated 04th April 2011 or later FAA approved revision. Installation of Custom Fiberglass Molding Skis, Model 3000-01(A/B), according to Master D (...) | Installer must check availability of EASA-approval for one or more referenced STC when airplane is modified per that / those STC(s). | 19.12.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|------------------|----------------|----------------------|---|---|---|--|---|---------------|
| 10014718 | CZECH AIRLINES TECHNICS A.S. | EASA.A.S.02895 | | | AIRBUS A320-214 | Cabin layout modification from 156 pax to 162 pax with seattype change /Removal of the existing seats for 156 passengers and doghouses, installation of new seats for 162 passengers, new doghouses, relocation and addition of PSU#s as required, instal (...) | | Modification No. M / 20 # 25 / 0002 | | 19.03.2007 |
| 10037313 | DAHER SOCATA A/S | | REV. 2 | | AS 355 F2 | Amendment of the configuration approved by STC 10037313, rev. 1 to addUHF (on version B only) and second radio altimeter indicator (onversions A and B), as described in "Design change description fornavigation system" DAHER SOCATA N° STC 34.001.11 E (...) | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | STC.34.001.11.002 FAR27 CS27 Airworthiness code compliance - Ed 1 Rev4- Design change description Ed 1 Rev 3 | Prior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noadverse effect upon the airworthiness ofthe produc (...) | 02.10.2013 |
| 10026805 | DANSK FLY ELEKTRONIK ApS | P-EASA.A.S.03944 | | | BEECH 1900D | Installation of Skytrac ISAT-100 GPS/iridium communicator system | As defined in the FAA TCDS A24CE | -Design Order Package DFE-DO-0361-DOP-00, dated 15-06-09-Declaration of Compliance, dated 11-07-09 | None | 11.08.2009 |
| 10027379 | DANSK FLY ELEKTRONIK ApS | P-EASA.A.S.03915 | | | PIPER PA-28-140 PIPER PA-28-150 PIPER PA-28-151 PIPER PA-28-160 PIPER PA-28-161 PIPER PA-28-180 PIPER PA-28-181 PIPER PA-28-235 PIPER PA-28-236 PIPER PA-28R-180 PIPER PA-28R-200 PIPER PA-28R-201 PIPER PA-28R-201T PIPER PA-28RT-201 PIPER PA-28RT-201T PIPER PA-28S-160 PIPER PA-28S-180 | Installation of ASPEN EFD1000 PRO Flight display | CS 23 Ammendment 1 | DFE-DO-0345 Complete Design Package | None | 25.09.2009 |
| 10027985 | DANSK FLY ELEKTRONIK ApS | | | | F172D, F172E, F172F, F172H, F172K, F172L, F172M, F172N, F172P | ASPEN EFD1000 Installation | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalCS-23The certificated noise and/ or emissions levels of the original productare unchang (...) | DFE Design order DFE-DO-00372 dated 01-09-2009 including AFMS.Instructions fro Continuing Airworthiness etc. | Not Applicable | 23.11.2009 |
| 10031092 | DANSK FLY ELEKTRONIK ApS | | | | PA-42, PA-42-720, PA-42-1000 | Installation of Avidyne EX500 MFD | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS-23The certificated noise and/ or emissions levels of the original productare unchan (...) | DFE Design Order DFE-DO-0375DFE-DO-0375 Design package Rev.00 or later revisions | None | 27.07.2010 |
| 10031093 | DANSK FLY ELEKTRONIK ApS | | | | PA-42, PA-42-720, PA-42-1000 | Installation of one or two Garmin GTX 330D Mode-S transponder | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS-23The certificated noise and/ or emissions levels of the original productare unchan (...) | DFE Design Order DFE-DO-0367DFE-DO-0367 Design Package Rev.00 or later revisions | None | 27.07.2010 |
| 10031129 | DANSK FLY ELEKTRONIK ApS | P-EASA.A.S.03981 | | | PA-42, PA-42-1000, PA-42-720 | Installation of dual Garmin GNS 430AW, dual Garmin GNS 530AW or GarminGNS 430AW and GNS 530AW | CS-23 | DFE-DO-0366 Data Package Rev. 00 or later revisions | For PA-42 with Bendix King EFIS only | 28.07.2010 |
| 10031404 | DANSK FLY ELEKTRONIK ApS | | | | B200, B200C | Installation of L-3 Skywatch HP Traffic Alert & Collision AvoidanceSystem, Model SKY899A. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalCS-23 Amendment 1The certificated noise and/ or emissions levels of the original produc (...) | DFE Design Order DFE-DO-0410 Design Package or later revisions of theabove listed documents approved by EASA | Garmin G1000 retrofit STC must be installed in aircraft prior toinstalling this STC. | 13.08.2010 |
| 10034209 | DANSK FLY ELEKTRONIK ApS | | | | B200 KING AIR | Installation of Garmin DUAL GPS 400. GPS 1 is linked to the autopilotssystem. Ref. DFE-DO-0401-PS-00 Product Specification. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalCS 23 Amendment 1The requirements for environmental protection and the associatedcerti (...) | Design Package DFE-DO-0401, including DFE-DD-0401#AFMS-00 Flight ManualSupplement Garmin GPS-400 and DFE-DD-0401-ICA-00Instructions forContinued Airworthiness or later revisions of the above listed documents approved by EASA | None | 16.03.2011 |
| 10034210 | DANSK FLY ELEKTRONIK ApS | | | | B200 KING AIR | Installation of Bendix/Kings KGP 560 EGPWS system with Avidyne EX500display. Ref. DFE-DO-0400-PS-00 Product Specification. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS 23 Amendment 1The requirements for environmental protection and the associatedcerti (...) | Design Package DFE-DO-0400, including DFE-DD-0400 #AFMS1-00FlightManual Supplement Avidyne EX-500, DFE-DD-0400 #AFMS2-00 Flight ManualSupplement EGPWS and DFE-DD-0400-ICA-00 Instructions for ContinuedAirworthiness or later revisionsof the above listed documents approved by EASA | None | 16.03.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------|-------------------|----------------|---------------------------|--|---|---|--|---|---------------|
| 10035243 | DANSK FLY ELEKTRONIK ApS | | | | CESSNA 414, 414A | Installation of Garmin GTX330D Transponder | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Design Order Package DFE-DO-0439 including Aircraft Flight Manual Supplement (DFE-DO-0439-AFMS) and Instructions for Continued Airworthiness (DFE-DO-0439-ICA) or later revisions of the above listed documents approved by EASA | None | 07.06.2011 |
| 10014207 | DAO AVIATION A/S | EASA.A.S.01358 | | | PIPER PA-24 PIPER PA-24-250 PIPER PA-24-260 PIPER PA-24-400 | Installation of Garmin GNS 430 COM/NAV/GPS. Approval for B-RNAV, en route IFR and Terminal flights | | DFE Design Order Package DO-0011. Garmin installation manual P/N: 190-00140-02 (Rev. L) SLV approved Aircraft Flight Manual supplement DFE-DD-AFMS-01-00. | | 29.06.2005 |
| 10014324 | DAO AVIATION A/S | EASA.A.S.01577 | | | BEECH 200, SUPER KING AIR | Installation of Honeywell ACAS II system comprising CAS 67A processor, 2 IVA-81D VSIs and 2 ANT-67A antennas. | | DFE Design order package DO-0059. Honeywell installation manual. Antenna installation design by Raytheon Aircraft Company Ref. no. JBB/0416-06. Drawing kit P/N 101-4092, rev 1. | MSN BB-309 only. | 27.10.2006 |
| 10014518 | DAO AVIATION A/S | EASA.A.S.02596 | | | | Avionics upgrade with following equipment (DFE-DO-0075-EPL-01 Equipment and Parts List, (Rev. 1) 1 GMA 347 Audio panel Garmin 1 GNS 330 COM/ NAV/ GPS Garmin 1 GNS 430 COM/ NAV/ GPS Garmin CDI | | DFE work order package DFE-DO-0075 and Installation Manuals according to Reference Document list, DFE-DO-75-ECO-01 page 1 and 2. Approved Flight Manual Supplements: DFE-DO-0075-AFMS-430 REV. IR, DFE-DO-0075-AFMS-530 REV. IR, DFE | | 13.07.2006 |
| 10044711 | DAO AVIATION A/S | | | | M20C, M20E, M20F, M20J, M20K | Installation of Electric Flight Display, Aspen EFD 1000 Pro. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | DAO Aviation A/S, Design order Package DAO-DO-0490-DOP-00, date 20-09-2012 DAO Aviation A/S, AFM-supplement DAO-DD-0490-AFMS-00, Rev. IR, date 20-09-2012 or later revisions of the above listed documents approved by EASA. | None | 30.04.2013 |
| 10016712 | DART AEROSPACE Ltd. | EASA.IM.R.S.00710 | REV. 1 | TCCA STC SH04-7 | BELL 205A BELL 205A-1 BELL 205B BELL 212 BELL 214B BELL 214B-1 BELL 412 BELL 412 EP | Installation of D412-702 Shoulder Harness Kits. Transport Canada STC No SH04-7 issue 1 dated 6th February 2004. The purpose of these kits is to add shoulder harnesses to existing seats that are only equipped with lap belts. | | ∫ MDL-D412-702 # Master Drawing List, Revision E dated 27-April-2007 or later approved revision; ∫ IIN-D412-702 # Installation Instruction, Revision D dated 12-April-2007 or later approved revision; ∫ ICA-D412-702 # Maintenance Manual Supplement (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 09.08.2007 |
| 10016723 | DART AEROSPACE Ltd. | EASA.IM.R.S.01021 | | SH04-31 | BELL 205A-1 BELL 205B BELL 212 BELL 214B BELL 214B-1 BELL 412 BELL 412 CF BELL 412 EP | Installation of Replacement Window Kits | | Installation must be in accordance with Dart Aerospace Installation Instruction IIN-D412-709rev A or later approved revision. Maintenance must be in accordance with Dart Aerospace Instruction for Continued Airworthiness ICA-D412-709 Rev A TCCA accepted or later approved revision | | 02.01.2006 |
| 10016748 | DART AEROSPACE Ltd. | EASA.IM.R.S.01074 | | TCCA STC SH04-30, ISSUE 2 | ROBINSON R22 ALPHA ROBINSON R22 BETA | Validation of TCCA STC SH04-30 for the installation of Cabin Floor Protectors. | | Installation Instructions IIN-D022-720, Revision A dated 6.04.2004 or later TCCA approved revisions. Instructions for Continued Airworthiness ICA-D022-720 Revision 0 or later TCCA accepted revisions. | 1. Installation and Maintenance in accordance with the above referenced Technical Documentation. 2. This STC is approved only for the product configuration as defined in the paragraph "Description". Compatibility (...) | 14.06.2005 |
| 10016749 | DART AEROSPACE Ltd. | EASA.IM.R.S.01075 | | TCCA STC SH04-40, ISSUE 1 | ROBINSON R44 ROBINSON R44 II | Validation of TCCA STC SH04-40 for the installation of Cabin Floor Protectors. | | Installation Instructions IIN-D044-719, Revision A dated 4.06.2004 or later TCCA approved revisions. Instructions for Continued Airworthiness ICA-D044-719 Revision 0 or later TCCA accepted revisions. | 1. Installation must be performed in accordance with Installation Instructions IIN-D044-719.2. Maintenance in accordance with Instructions for Continued Airworthiness ICA-D044-719.3. This STC is approved only for the product configuration as defined (...) | 14.06.2005 |
| 10016750 | DART AEROSPACE Ltd. | EASA.IM.R.S.01076 | | TCCA STC SH05-7, ISSUE 1 | ROBINSON R44 ROBINSON R44 II | Validation of TCCA STC SH05-7 for the installation of a Cold Weather Baffle Kit for Cold Weather operation of Robinson R44 and R44II. | | Installation Instructions IIN-D044-717 Rev. B TCCA Approved 25. 01.2005 or later TCCA approved revisions. Instructions for Continued Airworthiness ICA-D044-717 Rev. O TCCA accepted 01.02.2005 or later TCCA accepted revisions. | 1. Installation must be performed in accordance with Installation Instructions IIN-D044-717.2. Maintenance in accordance with Instructions for Continued Airworthiness ICA-D044-717.3. This STC is approved only for the product configuration as defined (...) | 14.06.2005 |
| 10016756 | DART AEROSPACE Ltd. | EASA.IM.R.S.01085 | | SH05-10, ISSUE 1 | BELL 205B BELL 212 BELL 214B BELL 214B-1 BELL 412 BELL 412 CF BELL 412 EP | Space door Installation | | Installation must be in accordance with Dart Aerospace Installation Instruction IIN-D412-694 rev A or later approved revision. Maintenance must be in accordance with Dart Aerospace instruction for Continued Airworthiness ICA-D412-694 Rev A TCCA accepted or later approved revision | | 02.01.2006 |
| 10016757 | DART AEROSPACE Ltd. | EASA.IM.R.S.01086 | | SH94-38, ISSUE 2 | SIKORSKY S-76A SIKORSKY S-76C | Installation of Cargo Mirrors | | Dart Aerospace Ltd. Drawing No IIN-D076-612 Revision A dated June 26, 2002 or later approved revision | | 25.01.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------|-------------------|----------------|-------------------------|---|--|---------------------|---|---|---------------|
| 10016758 | DART AEROSPACE Ltd. | EASA.IM.R.S.01087 | | TCCA STC SH05-14 | ROBINSON R22 ALPHA ROBINSON R22 BETA | Installation of Replacement Window Kits in accordance with TCCA STC SH05-14. | | Installation Instructions IIN-D022-714, Revision A approved 26.09.2004 or later approved revisions. Instructions for Continued Airworthiness ICA-D022-714 Revision 0 Transport Canada accepted 18.03.2005 or later Transport Canada accepted revisions. Flig (...) | 1. Installation, Operation and Maintenance must be held in accordance with the above referenced Technical Documentation. 2. This STC is approved only for the product configurations defined in the approved design data referred to in the paragraph "D (...) | 14.06.2005 |
| 10016759 | DART AEROSPACE Ltd. | EASA.IM.R.S.01088 | | TCCA STC SH05-13 | ROBINSON R44 ROBINSON R44 II | Installation of Replacement Window Kit in accordance with TCCA STC SH05-13. | | TCCA approved Flight Manual Supplement FMS-D044-713. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 13.06.2005 |
| 10016814 | DART AEROSPACE Ltd. | EASA.IM.R.S.01179 | | TCCA STC SH01-5 ISSUE 2 | BELL 206A BELL 206L BELL 206L-1 BELL 206L-3 BELL 206L-4 BELL 407 | Landing Gear Crosstubes | | ACP-D206-667 rev BACP-D407-667 rev AACR-D206-667-1 rev A | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 03.07.2006 |
| 10016815 | DART AEROSPACE Ltd. | EASA.IM.R.S.01180 | | SH04-54 ISSUE1 | BELL 204B BELL 205A BELL 205A-1 BELL 205B BELL 212 BELL 214B BELL 214B-1 BELL 412 | Installation of Fuel Purge Canister | | Installation Instructions: IIN-D412-706 Revision A, dated November 30, 2004 or later approved revision. Instructions for Continued Airworthiness: ICA-D412-706, dated November 30, 2004 or later approved revision. | | 26.04.2007 |
| 10016816 | DART AEROSPACE Ltd. | EASA.IM.R.S.01181 | | SH05-11 | BELL 204B BELL 205A BELL 205A-1 BELL 212 BELL 214B BELL 214B-1 BELL 412 BELL 412 EP | D212-722 Avionics Console | | ACR-D212-722-1 rev AICA-D212-722 rev 0 IIN-D212-722 rev ADR-D212-722 rev C | None | 30.06.2006 |
| 10016822 | DART AEROSPACE Ltd. | EASA.IM.R.S.01187 | | SH05-44 | BELL 206A BELL 206L-1 BELL 206L-3 BELL 206L-4 BELL 407 | D206/407-728 Skid tube wear plates. | | MDL-D407-728-Master Drawing List, Revision A dated 3 August 2005 or later approved revision; IIN-D407-728 # Installation Instruction, Revision A dated 3 August 2005 or later approved revision; ICA-D407-728 # Instructions for continued Airworthiness, Rev (...) | | 09.10.2007 |
| 10016828 | DART AEROSPACE Ltd. | EASA.IM.R.S.01198 | | SH03-6 | BELL 204B BELL 205A BELL 205A-1 BELL 212 BELL 214B BELL 214B-1 | D205-594 Extended height landing Gear TCCA STC SH03-6 | | Master Drawing List MDL #D205-594 Rev. A dated 6 July 2005 or later approved revision. Installation instruction IIN-D205-594 Rev. C dated 18 July 2005 or later approved revision. Instruction for continued airworthiness ICA-D205-594 Rev. 2 dated 18 Jul (...) | Maintenance of the Extended Height Landing Gear must be in accordance with Dart Aerospace Ltd. ICA-D205-594, Revision 2, dated 18 July 2005, or later approved revisions. Operation of the Bell 214B and 214B-1 model types with the Extended Height Land (...) | 03.10.2006 |
| 10016833 | DART AEROSPACE Ltd. | EASA.IM.R.S.01204 | | SH96-69 | BELL 205A BELL 205A-1 BELL 205B BELL 212 BELL 214B BELL 214B-1 BELL 412 BELL 412 EP | Installation of Heli Access Steps i.a.w. Transport Canada STC SH96-69, issue 7 | | - Flight Manual Supplement FMS-D412-630, Rev. D, dated 15.12.2005- Installation Instructions D412-630, Rev. J, dated 08.12.2005- Instructions for Continued Airworthiness ICA-D412-630, Rev. 1, dated 12.12.2005 | | 02.06.2006 |
| 10016866 | DART AEROSPACE Ltd. | EASA.IM.R.S.01242 | | TCCA STC SH96-120 | BELL 204 BELL 205A BELL 205A-1 BELL 205B BELL 212 BELL 214B BELL 214B-1 BELL 412 BELL 412 CF BELL 412 EP | D205-541 Heli-Utility Basket in accordance with Transport Canada STC SH96-120, Issue 3, dated May 16, 2006 | | - FMS-D412-541 for Bell 412 Models Revision A dated May 16, 2006 or later approved revision;- FMS-D205-541 for Bell 205/210/212 Models revision E dated May 16, 2006 or later approved revision;- FMS-D214-541 for Bell 214 models Revision A dat (...) | 1. VFR Only; 2. Doors to be closed at all times during flight; 3. Maximum Load 136 kg (300 lb) or 100kg (220lb) if middle support is not installed; 4. This STC is approved only for the product configuration as defined in the approved design (...) | 29.09.2006 |
| 10016876 | DART AEROSPACE Ltd. | EASA.IM.R.S.01253 | | TCCA STC SH04-17 | ROBINSON R44 | Installation of D044-715 Cargo Mirrors i.a. with TCCA STC SH04-17 | | - Installation Instructions IIN-D044-715, Rev. A, dated 20 01 2004.- Instructions for Continued Airworthiness ICA-D044-715, Rev. 0, dated 05.02.2004 | None | 11.12.2006 |
| 10016877 | DART AEROSPACE Ltd. | EASA.IM.R.S.01254 | | TCCA STC SH04-12 | ROBINSON R22 ALPHA ROBINSON R22 BETA | Installation of D022-716 Cargo Mirrors i.a. with TCCA STC SH04-12 | | - Installation Instructions IIN-D022-716, Rev. A, dated 20.01.2004.- Instructions for Continued Airworthiness ICA-D022-716, Rev. 1, dated 12.05.2004. | | 12.12.2006 |
| 10016901 | DART AEROSPACE Ltd. | EASA.IM.R.S.01278 | REV. 1 | | EUROCOPTER DEUTSCHLAND EC-135- | D135-751 Skidtube and Skidtube Extension Installation law FAA STC SR01709SE. | | Master Drawing List MDL-D135-751, Revision D Installation Instructions IIN-D135-751, Revision C Instructions for Continued Airworthiness ICA-D135-751, Revision 2 | | 10.12.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------|-------------------|----------------|--------------------------|--|--|---|--|--|---------------|
| 10016903 | DART AEROSPACE Ltd. | EASA.IM.R.S.01280 | REV. 2 | TCCA STC SH06-27 | AS350B, AS350B1, AS350B2 AS350B3, AS350BA, AS350D AS350E, AS350N, AS350NP AS350F, AS350F1, AS350F2 | Cross Tube Installation in accordance with Transport Canada STC SH06-27, Issue 1, dated 16 August 2006. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | MDL-D350-748 rev F dated 30 June 2010 - Master Document List-IIN-D350-748 rev B dated 15 June 2007 - Installation Instructions-ICA-D350-748 rev 1 dated 29 June 2010 - Instructions for Continued Airworthiness or later revisions of the above listed doc (...) | None | 03.09.2010 |
| 10016908 | DART AEROSPACE Ltd. | EASA.IM.R.S.01289 | REV. 1 | TCCA STC SH06-39 ISSUE 2 | AGUSTA A119, AW119 MKII | TCA STC No. SH06-39 Issue 2 - "Helicopter Utility Basket Installation" - This installation consists in a box, fixed to the landing gear, for goods transportation. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCA STC No. SH06-39 Issue 2 dated 2 July 2009 FMS-D119-756-EASA Rev. A, ICA-D119-756-EASA Rev. 0, MDL-D119-756-EASA Rev. A or later revisions of the above listed documents approved by EASA or later revisions of the above listed documents approved by EAS (...) | None | 10.02.2011 |
| 10016909 | DART AEROSPACE Ltd. | EASA.IM.R.S.01290 | | TCCA STC SH93-97 | AS 350 B, AS 350 B1, AS 350 B2 , AS 350 B3, AS 350 BA , AS 350 D, AS 350 D1, AS350E, ES350F, AS350F1, AS350F2, AS350N, AS350NP | D350-602 Helicopter Utility Pod # Validation of Transport Canada STC SH93-97. | | Compliance Report - ACR-D350-602-1 Rev A Instructions for Continued Airworthiness - ICA-D350-602 Rev 0 or later approved revision Flight Manual Supplement - FMS-D350-602 Rev C or later approved revision Master Document List - MDL-D350-602 Rev A or later (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. AS355 (...) | 02.06.2008 |
| 10016911 | DART AEROSPACE Ltd. | EASA.IM.R.S.01292 | REV. 1 | TCCA STC SH93-88 | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 EUROCOPTER FRANCE AS 350E EUROCOPTER FRANCE AS 350F EUROCOPTER FRANCE AS 350F1 EUROCOPTER FRANCE AS 350F2 EUROCOPTER FRANCE AS 350N | Spacepod Installations (Validation of Canadian STC Number SH93-88) | | D350-600 Installation Manual : IIN-D350-600, Revision B Maintenance Instructions : ICA-D350-600, Revision 3 Flight Manual Supplement : FMS-D355-600, Revision D | | 09.04.2008 |
| 10016918 | DART AEROSPACE Ltd. | EASA.IM.R.S.01303 | | TCCA STC SH96-88 | BELL 204B BELL 205A BELL 205A-1 BELL 205B BELL 212 BELL 214B BELL 214B-1 BELL 412 BELL 412 EP | Skidtube Installation | | # For all interested products: MDL-D205-634 Revision A dated 13 April 2006 or later approved revision - Master Document List # Bell 204D204-635 Revision C dated 30 March 2006 or later approved revision - Installation Instructions. ICA-D204-635 Re (...) | 1. Operation of the Bell 214B and 214B-1 model types with Dart D205-634-011 skidtubes and the required Dart D212-664-101/201 Crosstube installation as incorporated by EASA STC No. IM.R.S.01304 (Canadian STC SH01-9) is limited to a maximum take of (...) | 27.06.2007 |
| 10016919 | DART AEROSPACE Ltd. | EASA.IM.R.S.01304 | | TCCA STC SH01-9 | BELL 204B BELL 205A BELL 205A-1 BELL 205B BELL 212 BELL 214B BELL 214B-1 BELL 412 BELL 412 CF BELL 412 EP | Cross Tube Installation in accordance with Transport Canada STC SH01-9, Issue 3, dated 8 June 2005, Revision to LBA STCRC 1181 dated 15 January 2002 | | For Bell 214B and 214 B-1 models only FMS-D214-664 Revision A dated 8 June 2005 or later approved revision; Master Document List MDL-D212-664 revision A dated 8 June 2005 or later approved revision; Installation Instructions IIN-D212-664 revision 2 dat (...) | Bell 214B and 214B-1 models only Operation of the Bell 214B and 214B-1 model types with the cross tube installation is limited to a maximum take off and landing weight of 5403 kg (11,900 lbs). Longitudinal centre of gravity limits have changed accord (...) | 06.03.2007 |
| 10016942 | DART AEROSPACE Ltd. | EASA.IM.R.S.01335 | | TCCA STC SH02-8; | ROBINSON R44 ROBINSON R44 II | D044-662 Bearpaws. Revision to DGAC STC IM225-SI-0026, DGAC approved 30/04/2003, with addition of R44II aircraft model and larger Bearpaw Kits. | | Dart Aerospace Ltd. Installation Instructions IIN-D044-662 rev. D, dated 14/02/2007 or later approved revisions. Dart Aerospace Ltd. Instruction for Continued Airworthiness ICA-D044-662, Rev. 1 dated 14/02/2007 or later approved revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and #Associated Technical Documentation#. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 31.05.2007 |
| 10016962 | DART AEROSPACE Ltd. | EASA.IM.R.S.01361 | | TCCA STC SH07-37 | BELL 205A BELL 205A-1 BELL 205B BELL 212 BELL 412 BELL 412 CF BELL 412 EP | D412-760 Cabin Floor Protectors. | | MDL-D412-760 # Master Drawing List, Revision A dated 29 June 2007 or later approved revision; IIN-D412-760 # Installation Instruction, Revision A dated 8 June 2007 or later approved revision; ICA-D412-760 # Maintenance Manual Supplement, Revision 0 dated 29 June 2007 or later approved revision | | 06.09.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------|-------------------|----------------|--------------------------------|--|---|---|--|--|---------------|
| 10016984 | DART AEROSPACE Ltd. | EASA.IM.R.S.01389 | REV. 1 | TCCA STC SH93-4 | SEE EASA APPROVED MODEL LIST | Bearpaw Installation | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AS 350 and AS 355 models: MDL-D350-578 revision A dated 11 June 2010 - Master Document List; D350-578 revision F dated 29 August 2008 - Installation Instructions; ICA-D350-578 revision 1 dated 28 August 2008 - Instructions for Continued Airworthiness. EC1 (...) | none | 22.09.2010 |
| 10016991 | DART AEROSPACE Ltd. | EASA.IM.R.S.01398 | | TCCA STC SH92-20, ISSUE 2; FAA | BELL 204B BELL 205A BELL 205A-1 | D204-575/D205-576 Cabin Heating Systems, Transport Canada STC SH92-20 Issue 2 dated 14 May 1992 | | Bell 205A and 205A-1 aircraft: Dart Aerospace Drawing 205-576 Rev D dated 21 August 2008; Dart Aerospace Installation Instructions IIN-D205-576 Rev D dated 21 August 2008; Flight Manual Supplement No FMS D205-576 Rev D dated 21 August 2008; Maintenance Ma (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 03.09.2008 |
| 10016996 | DART AEROSPACE Ltd. | EASA.IM.R.S.01405 | REV. 2 | TCCA STC SH94-14 | 1. AS350B/B1/B2/B3/BA/D/D1 1. EC130B4 2. AS355E/F/F1/F2/N/NP | D350-607, D130-701 Basket Installations This revision adds the models AS355N and NP. | For AS350/355 models: FAR 27 up to and including Amendment 27-23 for the affected areas. For EC130B4 model: FAR 27 up to and including Amendment 27-32 for the affected areas. The requirements for environmental protection and the associated certificated n (...) | AS350/355 models: FMS-D350-607 revision E dated 10 December 2010 - AS350 Flight Manual Supplement; FMS-D355-607 revision E dated 10 December 2010 - AS355 Flight Manual Supplement. EC130B4 Model: FMS-D130-701 revision D dated 10 December 2010 - EC130 Flight (...) | None | 24.01.2011 |
| 10017030 | DART AEROSPACE Ltd. | EASA.IM.R.S.01449 | REV. 1 | TC STC SH05-45 | BELL 206A BELL 206B | D206-746 Shroud Air Flow Heater System. | | IIN-D206-746 Revision A dated 16 May 2006 or later approved revision # Installation Instructions; MDL-D206-746 Revision G dated 11 February 2008 or later approved revision # Master Document List; MMS-D206-746 Revision A dated 16 March 2006 or later approved revision # (...) | | 16.04.2008 |
| 10017031 | DART AEROSPACE Ltd. | EASA.IM.R.S.01451 | | TCCA STC SH99-221 ISSUE 1 | EUROCOPTER FRANCE EC 120 B | D120 -638 Bearpaw Installation. | | z Definition and Installation in accordance with: Dart Aerospace Ltd. Installation instructions IIN-D120-638, revision A, dated July 7, 1999, or later approved revision. z Inspection and Maintenance in accordance with: Dart Aerospace Ltd. Maint (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 13.08.2008 |
| 10017032 | DART AEROSPACE Ltd. | EASA.IM.R.S.01452 | | TCCA STC SH01-25 ISSUE 1 | EUROCOPTER FRANCE EC 120 B | D120 -639 Wearplate Installation. | | z Definition and Installation in accordance with: Dart Aerospace Ltd. Installation instructions IIN-D120-639, revision A, dated December 11, 2000, or later approved revision. z Inspection and Maintenance in accordance with: Dart Aerospace Ltd. In (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 13.08.2008 |
| 10017033 | DART AEROSPACE Ltd. | EASA.IM.R.S.01456 | | TCCA STC SH08-5 | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 EUROCOPTER FRANCE AS 355N EUROCOPTER FRANCE AS 355NP | D350-766 interior Trim Kit Installations | | IIN-D350-766 revision A dated 3 March 2008 or later approved revision # Installation Instructions; MDL-D350-766 revision A dated 3 March 2008 or later approved revision # Master Document List; ICA-D350-766 revision 0 dated 3 March 2008 or later approved revision (...) | | 30.04.2008 |
| 10017059 | DART AEROSPACE Ltd. | EASA.IM.R.S.01492 | REV. 1 | TCCA STC SH08-40 | BELL 212 BELL 412 BELL 412 EP | Replacement Goggles Installation. | | - IIN-D412-776-Rev. B, INSTALLATION INSTRUCTIONS or later approved revision; MDL-D412-776-Rev. B, MASTER DOCUMENT LIST or later approved revision; ICA-D412-776-Rev. 0, INSTRUCTIONS FOR CONTINUED AIRWORTHINESS or later approved revision | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. (...) | 04.05.2009 |
| 10017074 | DART AEROSPACE Ltd. | EASA.IM.R.S.01516 | REV. 1 | TCCA STC SH91-20 | AS350 AS350B1, AS350B2 AS350B3, AS350BA, AS350C AS350D, AS350D1 AS355E, AS355F, AS355F1 AS355F2, AS355N, AS355NP | D350-561 Cargo Mirror installations. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | MDL-D350-561 revision A dated 16 June 2010 - Master Document List; D350-561 revision D dated 16 June 2010 - Installation Instructions; ICA-D350-561 revision 0 dated 16 June 2010 - Instructions for Continued Airworthiness or later revisions of the above I (...) | none | 03.09.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------|---------------------|----------------|--------------------------|--|---|--|--|---|---------------|
| 10017075 | DART AEROSPACE Ltd. | EASA.IM.R.S.01518 | | TC STC SH08-37 ISSUE 1 | BELL 205A-1 BELL 205B BELL 212 BELL 214B BELL 214B-1 | D205-778 Comfort Seat Installation | | MDL-D205-778 revision A dated 9 September 2008 or later EASA approved revision # Master Document List:IN-D205-778 revision A dated 25 April 2008 or later EASA approved revision # Installation Instructions:ICA-D205-778 revision 0 dated 25 April 2008 o (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 09.01.2009 |
| 10017077 | DART AEROSPACE Ltd. | EASA.IM.R.S.01526 | REV. 2 | TCCA STC SH92-16 | 1. 204B, 205A-1 2. 212, 412, 412EP 3. 214B, 214B-1 | D212-580 Cargo Mirror/LED Pulse Light Installation This revision adds the LED Pulse light installation to the Cargo Mirrors. | FAR 29 up to and including Amendment 29-51 for the affected areas. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | MDL-D212-580 revision D dated 13 August 2010 - Master Document List:IN-D21-580 revision D dated 8 February 2010 - Installation Instructions:ICA-D212-580 revision 2 dated 20 July 2010 - Instructions for Continued Airworthiness; or later revisions of (...) | None | 26.11.2010 |
| 10017078 | DART AEROSPACE Ltd. | EASA.IM.R.S.01528 | | TCCA STC SH08-08 | BELL 205A BELL 205A-1 | Replacement Engine Mount Installation (kit 205-770) | | IN-D205-770 revision A dated 1 November 2008 or later EASA approved revision # Installation Instructions: MDL-D205-770 revision B dated 24 November 2008 or later EASA approved revision # Master Document List: ICA-D205-770 revision 0 dat (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 26.02.2009 |
| 10017079 | DART AEROSPACE Ltd. | EASA.IM.R.S.01529 | | TCCA STC SH08-53 | AGUSTA AB-412 AGUSTA AB-412EP BELL 204B BELL 205A-1 BELL 205B BELL 212 BELL 214B BELL 214B-1 BELL 412 BELL 412 EP | Procedure Trim Installations (kit D412-775). | | IN-D412-775 revision A dated 14 November 2008 or later EASA approved revision # Installation Instructions:MDL-D412-775 revision A dated 1 December 2008 or later EASA approved revision # Master Document List:ICA-D412-775 revision 0 dated 14 November (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 25.02.2009 |
| 10027063 | DART AEROSPACE Ltd. | P-EASA.IM.R.S.01493 | REV. 1 | TCCA STC SH08-38 | 205A-1, 205B, 212, 412, 412EP | SH08-38 Issue 1 "D412-769 Air Facing Seat Installation" | FAR 29 | - Installation Instructions: IIN-D412-769 Rev. B dated 24 June 2008-, Master Document/ Master Drawing List, MDL-D412-769 Rev. B dated 5 June 2008-, Instructions for Continued Airworthiness, ICA-D412-769 Rev. 0 dated 24 June 2008 | For Instructions for Continued Airworthiness see Report ICA-D412-769 Rev. 0 dated 24 June 2008. | 05.11.2009 |
| 10028823 | DART AEROSPACE Ltd. | P-EASA.IM.R.S.01554 | | TCCA STC SH09-16 | EC130B4 | Traffic Advisory System Installation. | The original product certification basis as per EASA TCDS.R.009 Issue 03, 12 October 2007. The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original (...) | Definition and installation: DART Aerospace Ltd, Master Document List MDL-D130-787, Rev. B dated April 20, 2009; DART Aerospace Ltd, Installation Instructions IIN-D130-787, Rev. B dated April 06, 2009; Inspection and maintenance: Inspections for Continued A (...) | None. | 12.02.2010 |
| 10028917 | DART AEROSPACE Ltd. | P-EASA.IM.R.S.01556 | | TCCA STC SH09-04 | BELL 204B, 205A-1, BELL 212, 214B, 214B-1, BELL 412, 412EP | "Full Vertical Reference Door for Left Hand Vertical Reference Operations" | As defined on FAA TCDS H4SW and H1SW The Certification Basis for the original product remains applicable to this certificate/ approval. | -FMS-D205-761 Rev. A dated 26.01.2009 and TCCA approved 02.02.2009-FMS-D412-761 Rev. A dated 29.01.2009 and TCCA approved 02.02.2009-FMS-D214-761 Rev. A dated 29.01.2009 and TCCA approved 02.02.2009 or later revisions of the above listed documents ap (...) | Full compliance has not been demonstrated for airworthiness standards 29.1303(g)(h)(i)(j), 29.1321(b) and 29.1523(b) for single pilot operation from the Left Seat resulting in a Restricted Category classification. When conducting vertical reference work (...) | 10.03.2010 |
| 10029207 | DART AEROSPACE Ltd. | P-EASA.IM.R.S.01485 | | TCCA STC SH99-11 ISSUE 5 | 204B, 205A-1 | "Adjustable Engine Mount Kit" | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | MDL-D205-708 Rev. C Master Document List-IN-D205-708 Rev.A Installation Instructions-ICA-D205-708 Rev. 0 Instructions for Continued Airworthiness | None | 11.03.2010 |
| 10029712 | DART AEROSPACE Ltd. | P-EASA.IM.R.S.01380 | | TCCA STC SH07-36 | A119 & AW119 MKII | TCA STC SH07-36 Issue 2 "Replacement Windows Kit" - D119-755-011/14 - This STC involves removing of the existing window and installing a flat window (-011) in LH or RH crew door installing a bubble window (-14) in the RH crew door) | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FMS - D119-755 Rev. Cor later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision) | Not Applicable | 15.04.2010 |
| 10031514 | DART AEROSPACE Ltd. | | | TCCA STC SH05-4 | AS350 B, B1, B2, B3, BA, D | Installation of an Auxiliary Side Locker Fuel Tank. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List MDL-D350-794, revision D, dated 09.08.2010 - Flight Manual Supplement FMS-D350-794, revision C, dated 09.08.2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or (...) | none | 20.08.2010 |
| 10032253 | DART AEROSPACE Ltd. | | | TCCA STC SH09-37 ISSUE 1 | 1. 214B, 214B-1 2. 212, 412, 412 EP 3. 204B, 205A-1 | D412-791 Safety Enhancement kits | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FMS-D205-791 rev. A, FMS-D412-791 rev. A, FMS-D212-791 rev. A, ICAD-412-791 rev. 0, MDL-D412-791 rev. A, IIN-D412-791 rev. B or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision) | None | 05.11.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------|-----------------------|----------------|------------------------------|---|--|---|--|--|---------------|
| 1003253 | DART AEROSPACE Ltd. | | | TCCA STC SH10-44 | 206, A, A1, B, B1 L, L1, L3, L4, 407 | Automatic Door Opener Kits (Cabin and Baggage Doors). | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 27 amendment 27-45 The requirements for environmental protection and the associated cert (...) | MDL-D407-802 revision A dated 17 November 2010 - Master Document List: IN-D407-802 revision A dated 13 October 2010 - Installation Instructions; ICA-D407-802 revision 0 dated 13 October 2010 - Instructions for Continued Airworthiness. | None | 07.01.2011 |
| 10034340 | DART AEROSPACE Ltd. | | | TCCA STC SH10-49 | 429 | Installation of UltraPaws. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS27/29 Change 2 The requirements for environmental protection and the associated cert (...) | MDL-D429-793 revision A dated 27 October 2010 - Master Document List: IN-D429-793 revision B dated 25 October 2010 - Installation Instructions; ICA-D429-793 revision 0 dated 19 October 2010 - Instructions for Continued Airworthiness. | None | 25.03.2011 |
| 10034634 | DART AEROSPACE Ltd. | EASA.IM.R.S.01560 | | TCCA STC SH92-18 | AS350B, BA, B1, B2, B3, D, D1 | D350-567 Vertical Reference Window Installations. Addition of AS350DI to list of models. This STC revises and supersedes DGAC France STC 84-5[1103, dated 27 September 2002. | | - MDL-D350-567 revision A dated 23 January 2009 or later approved revision - Master Document List; - D350-567 revision D dated 8 July 2008 or later approved revision - Installation Instructions; - ICA-D350-567 revision 0 dated 20 January 2009 or later a (...) | None | 17.07.2009 |
| 10034636 | DART AEROSPACE Ltd. | EASA.IM.R.S.01561 | | TCCA STC SH92-18 | AS355E, F, F1, F2, N, NP | D350-567 Vertical Reference Window Installations. This STC is a revision of and supersedes DGAC-F STC 168-511104 dated 27 September 2002 for addition of AS355N and NP to list of models. | | - MDL-D350-567 revision A dated 23 January 2009 or later approved revision - Master Document List; - D350-567 revision D dated 8 July 2008 or later approved revision - Installation Instructions - ICA-D350-567 revision 0 dated 20 January 2009 or later ap (...) | None | 31.07.2009 |
| 10034637 | DART AEROSPACE Ltd. | EASA.IM.R.S.01453 | | TCCA STC SH02-33 | 1. AS350 D, B, B1, B2, BA, B3 1. (OP-3369 OPTION EXCLUDED) 2. AS355 E, F, F1, F2, N | Installation of Dual High Back Seat, Left Forward Position (Modification approved by TCCA under STC SH02-33 issue 1) | | - Dart Aerospace Limited Installation Instructions IIN-D350-689, rev Adated May 30, 2001 or later EASA approved revisions. - Flight Manual Supplement FMS-D350-689, rev A dated June 6, 2001 or later EASA approved revisions. - Instructions for Continue (...) | None | 03.06.2009 |
| 10035429 | DART AEROSPACE Ltd. | | REV. 2 | TCCA STC SH11-1 | 206L, 206L-1, 206L-3, 206L-4 407 | D407-797 Heli Utility Basket and Heli Access Step Installation, Revision to STC 10035429 to add D407-797-311 Heli Access Step. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The applicant complied voluntarily with the following paragraphs of FAR 27 amendment 27 (...) | FMS-D407-797 revision A dated 12 January 2011 - Flight Manual Supplement or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision). IIN-D407-797 revision C (...) | None | 28.11.2011 |
| 10039042 | DART AEROSPACE Ltd. | | REV. 1 | | 206 A/B/L/L1/L3/L4 | Screw In Windshield Modification. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Drawing List MDL-D206-817, Rev A, Dated February 06, 2012; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision). - Approved Dart Aerospace Lt (...) | None | 30.04.2012 |
| 10042362 | DART AEROSPACE Ltd. | DGAC STC 168-SI0994 | REV. 1 | TCCA SH 90-4 ISSUE 4 | AS335E, AS335F, AS335F1, AS335F2, AS335N, AS335NP. | Installation of rear locker extender as defined in Dart Aerospace Ltd Master Document List No. MDL-D350-604 Rev. C. Note 1. This approval amends DGAC STC 168-SI0994 to add the AS335N and AS335NP aircraft models. Note 2. This STC is the validation of TCCA STC SH90-4 Issue 4. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FMS D350-604 Rear Locker Rev. B dated 15/11/2011 or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU /Canada Bilateral Agreement. | None | 27.11.2012 |
| 10042575 | DART AEROSPACE Ltd. | LBA STC RC 1019 | REV. 1 | TCCA STC SH96-61 ISSUE NO. 4 | 1. AS350B, BA, B1, B2, B3, D 2. AS355E, F, F1, F2 | Installation of Left Hand and/or Right Hand Cargo Mirrors as defined in Eurocopter Canada Limited Master Drawing List MDL-96-005. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ECL Master Drawing List No. MDL096-005, Issue FECL. Instructions for Continued Airworthiness No. ICA-ECL-131, Rev. 0 | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 17.09.2013 |
| 10042994 | DART AEROSPACE Ltd. | DGAC-F STC 168-SI0996 | | TCCA STC SH 96-10 REV. 3 | AS335 E/F/F1/F2/N/NP | Emergency Litter Installations Installation of an Emergency Litter Kit as defined in Dart Aerospace Ltd. Master Document List MDL-D350-616. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: FAR 27 including amendments 27-1 through 27-20, 27.807, 27.1585 and 27.1587 at amendment 27-21 The requirements for environ (...) | Flight Manual Supplement FMS-D350-616 revision B or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/ Canada Bilateral Agreement. - Installation Instructions IIN-D350-616 Rev. (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 03.07.2013 |
| 10043041 | DART AEROSPACE Ltd. | P-EASA.IM.R.S.01541 | | TCCA SH 08-60 | 206, A, A-1, B, B-1, 206, L, L-1, L-3, L-4, 407 | D407-781 Interior Trim Panel Installations. | | - IIN-D407-781 revision A dated 22 October 2008 or later approved revision - Installation Instructions - MDL-D407-781 revision A dated 19 December 2008 or later approved revision - Master Document List; - ICA-D407-781 revision 0 dated 20 November 2008 (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The (...) | 05.06.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------|-------------------|----------------|----------------------|---|---|---|--|--|---------------|
| 10045158 | DART AEROSPACE Ltd. | | REV. 1 | TCCA SH12-54 ISSUE 1 | AW139 | AB139/AW139 Utility Interior Kits Installation | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Document List document MDL-D139-816 Revision A- Instructions for Continued Airworthiness document ICA-D139-816 Revision for later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 06.06.2013 |
| 10045487 | DART AEROSPACE Ltd. | | | TCCA STC SH13-18 | MBK-BK117 A1/A3/A4/B1/B2/C1/C2 | D117-840 Bearpaw Installation Installation of Bearpaw as defined in Master Document List MDL-D117-840. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Document List MDL-D117-840 Rev. A- Installation Instructions IIN-D117-840 Rev. A- Instructions for Continued Airworthiness ICA- D117-840 Rev. 0 | None | 28.06.2013 |
| 10016785 | DART AEROSPACE USA, Inc. | EASA.IM.R.S.01126 | | FAA STC SR01583SE | AGUSTA AB-412 AGUSTA AB-412EP BELL 205A BELL 205A-1 BELL 205B BELL 212 BELL 412 BELL 412 CF BELL 412 EP | Installation of DART D412-742 Float Skid tubes (FAA STC SR01583SE issued on 29 July 2005) | | - Master Drawing List MDL-D412-742, Rev. D dated 27 January 2006 or later approved revision- Airworthiness Compliance Document ACP-D412-742-1 Rev. B dated 26 May 2005- Instruction for continued airworthiness ICA-D412-742 Rev. 1 dated 3 October 2005 or later approved revision | Installation Instruction IIN-D412-742 Rev. B dated 3 October 2005 or later approved document. This approval should not be extended to other rotorcraft of this model on which other previously approved modifications are incorporated unless it is determined (...) | 12.05.2006 |
| 10016934 | DART AEROSPACE USA, Inc. | EASA.IM.R.S.01320 | REV. 1 | FAA STC SA00475SE | BELL 206 BELL 206A BELL 206B BELL 206L BELL 206L-1 BELL 206L-3 BELL 206L-4 BELL 407 | Replacement of the original landing gear skid tubes and saddles with DART Aerospace Model D206-642 skid tubes and saddles. This TV modifies STC EASA.IM.R.S.01320 | | MDL-D206-642 revision Q dated 8 March 2007 or later approved revision # Master Drawing List; D206-642 revision K dated 25 September 2006 or later approved revision # Installation Instructions; ICA-D206-642 revision O dated 7 March 2007 or later approved revision (...) | | 10.09.2007 |
| 10016945 | DART AEROSPACE USA, Inc. | EASA.IM.R.S.01339 | | TCCA STC SH98-14 | BELL 206A BELL 206B BELL 206L BELL 407 | D206-628 Heli Access Steps. This STC has been classified as #BASIC# by the FAA and is approved under the Type Validation Principles. | | MDL-D206-628 # Master Drawing List, Revision A dated 10 December 1997 or later approved revision; IIN-D206-628 # Installation Instruction, Revision A dated 28 November 1997 or later approved revision; MMS-D206-628 # Maintenance Manual Supplement, Rev. 1 (...) | | 26.07.2007 |
| 10016972 | DART AEROSPACE USA, Inc. | EASA.IM.R.S.01373 | REV. 1 | FAA STC SH4748NM | AS350B, AS350B1, AS350B2, AS350B3, AS350BA, AS350D. | Installation of a replacement instrument panel visor. This STC is a validation of FAA STC SH4748NM | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Geneva Aviation Master Drawing List, doc. ref. GA127, Revision D, dated March 8, 2008 Installation Instructions, doc. ref. GA127-3, Revision B, dated March 8, 2008 Compliance Checklist, doc. ref. CCL-127, Revision -, dated August 17, 2007 or later revisions of the above listed documents. | None | 15.04.2013 |
| 10017001 | DART AEROSPACE USA, Inc. | EASA.IM.R.S.01412 | | FAA STC SR00211SE | EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 C EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 EUROCOPTER FRANCE AS 350B EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 EUROCOPTER FRANCE AS 355N | D350-764 Shoulder Harness Kits. This STC has been classified as "BASIC" by the FAA and has been approved under the PTVP. | | GA 159 Revision A dated 14 June 1995 or later approved revision- Geneva Aviation Master Drawing List; MDL-D350-764 Revision C dated 18 December 2007 or later approved revision - Dart Aerospace USA Master Document List; IIN-D350-764 revision B dated 16 O (...) | | 05.03.2008 |
| 10017003 | DART AEROSPACE USA, Inc. | EASA.IM.R.S.01414 | REV. 1 | FAA STC SH4785NM | AS350B, AS350B1, AS350B2, AS350B3, AS350BA, AS350D. | Installation of replacement dome light/air vent assembly. This STC is a validation of FAA STC SH4785NM. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Geneva Aviation Master Drawing List, doc. ref. GA129, Revision B, dated September 22, 2011 Installation Instructions, doc. ref. GA129-2, Revision C, dated September 22, 2011 Compliance Checklist, doc. ref. CCL-129, Revision B, dated 15 February, 2008 Inst (...) | None | 15.04.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------|------------------------|----------------|----------------------|---|--|--|--|---|---------------|
| 10017034 | DART AEROSPACE USA, Inc. | EASA.IM.R.S.01458 | | FAA STC SR00948SE | AGUSTA A-119 AGUSTA AW-119 MKII | FAA STC SR00948SE #D119-675 Helicopter Access-Step#. | | Drawin List D119-675 Rev. D Dart Aerospace Installation Instructions D119-675 Rev. B Instruction for Continued Airworthiness D119-675 Rev. 1 | | 18.04.2008 |
| 10017045 | DART AEROSPACE USA, Inc. | EASA.IM.R.S.01469 | REV. 1 | FAA STC SR00910SE | EUROCOPTER CANADA BO-105 LS A- EUROCOPTER DEUTSCHLAND BO-105 | D105-674 Skidtube Installation i.a. with FAA STC SR00910SE | | - Dart Master Drawing List MDL-D105-674, Rev B, dated 1st June 2005 or later approved revisions- Dart Instructions for Continued Airworthiness ICA-D105-674, dated 17th March 2000 or later approved revisions | | 07.01.2009 |
| 10017052 | DART AEROSPACE USA, Inc. | EASA.IM.R.S.01477 | | FAA STC SR00914SE | EUROCOPTER FRANCE SA 315B | Installation of Dart Bearpaws P/N D315-670-011 - Validation of FAA STC SR00914SE. | | DART Aerospace Drawing List MDL-D315-670, revision B, dated 1-June-2005, or later approved revisions Dart Aerospace Installation Instructions IIN-D315-670, revision C, dated 4-April-2000, or later approved revisions. Dart Aerospace Maintenance Manual M (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 06.02.2009 |
| 10017053 | DART AEROSPACE USA, Inc. | EASA.IM.R.S.01478 | | FAA STC SR00848SE | EUROCOPTER FRANCE SA 315B | Installation of Dart Aerospace Skidtube P/N D315-668-011 (LH) and -012 (RH). Validation of FAA STC SR00848SE. | | DART Aerospace Installation Instructions IIN-D315-668, Revision A, dated 20-Sept-1999, or later approved revisions. DART Aerospace Maintenance Manual Supplement MMS-D315-668, Revision A, dated 13-Sept-1999, or later approved revisions. DART Aerospace M (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 06.02.2009 |
| 10017063 | DART AEROSPACE USA, Inc. | EASA.IM.R.S.01499 | REV. 1 | FAA STC SR01853SE | MBB-BK117 A-1/A-3/A-4 MBB-BK117 B-1/B-2/C-1/C-2 | D117-762 Skidtube installation Installation of skid tubes as defined in Master Drawing List MDL-D117-762. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List MDL-D117-762, Rev. D, dated July 2, 2013 Installation Instructions IIN-D117-762, Revision B, dated January 24, 2013 Instructions for Continued Airworthiness ICA-D117-762, Revision 1, dated February 13, 2013 | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 16.10.2013 |
| 10017084 | DART AEROSPACE USA, Inc. | EASA.IM.R.S.01551 | | FAA STC SR01043SE | AGUSTA A119 | Cargo Mirror Installation. Note: This STC supersedes ENAC STC N-CS-014 being the validation of FAA STC SR01043SE dated 25 January 2002. | | - IIN-D119-696 Rev. A Installation Instructions or later approved revision- MDL-D119-696 Rev. C Master Document List or later approved revision- ICA-D119-696 Rev. 0 Instructions for Continued Airworthiness or later approved revision- FMS (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 11.05.2009 |
| 10029107 | DART AEROSPACE USA, Inc. | N-CS-022 | | FAA STC SR00949SE | A119, AW119MKII | FAA STC SR00949SE dated 21 August 2008 'Bearpaw Installation' | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | IIN-D119-647 Rev. A Installation Instructions MDL-D119-647 Rev. C Master Document List- ICA-D119-647 Rev. 0 Instructions for Continued Airworthiness or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 05.03.2010 |
| 10033942 | DART AEROSPACE USA, Inc. | CAA UK 28081 AND 28082 | | FAA STC SR00646SE | 1. AS350 B, AS350 B1, AS350 B2 1. AS350 B3, AS350 BA, AS350 D 2. AS355 E, AS355 F, AS355 F1 2. AS355 F2, AS355 N, AS355 NP | D350-636 Skidtube Installations Note: This STC supersedes CAA Approval Notes 28081 and 28082 and adds models AS355N and NP to applicability. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | MDL-D350-636 revision N dated 1 September 2010 - Master Document List- IIN-D350-636 revision H dated 26 July 2010 - Installation Instructions- ICA-D350-636 revision 1 dated 26 July 2010 - Instructions for Continued Airworthiness. | None | 21.02.2011 |
| 10038103 | DART AEROSPACE USA, Inc. | | REV. 1 | FAA STC SR01042SE | EC135 P1(CDS) EC135 P1(CPDS) EC135 P2(CPDS), EC135 P2+ EC135 T1(CDS), EC135 T1(CPDS) EC135 T2(CPDS), EC135 T2+ | Dart D135-692-011 Bearpaws in accordance with Dart Aerospace Drawing List No. MDL-D135-692, and installation of this system. This EASA STC is based on validation of FAA STC SR01042SE law the PTVP. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement FMS-D135-692- EASA Revision A or later revisions of the above listed documents approved by EASA- Dart Aerospace Drawing List No. MDL-D135-692, Revision A, dated 9 October 2001- Installation Instruction IIN-D135-692, Revision (...) | Approval of this change in type design applies only to Eurocopter EC135 rotorcraft, which are equipped with standard skidtubes, high gear skidtubes, or float skidtubes as defined in the installation instructions. | 27.01.2012 |
| 10038656 | DART AEROSPACE USA, Inc. | | REV. 1 | FAA STC SR02138SE | AW139 | EASA Validation of FAA STC SR02138SE 'Installation of Dart D139-799 Maintenance Kits'. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Dart Master Document List MDL-D139-799, Revision C, dated June 6, 2010; Dart Installation Instructions IIN-D139-799, Revision D, dated June 10, 2011; Dart Instructions for Continued Airworthiness ICA-D139-799, Revision 3, dated May 5, 2011; or later rev (...) | None | 19.03.2012 |
| 10040294 | DART AEROSPACE USA, Inc. | | REV. 1 | FAA STC SR02024SE | A119, AW119MKII | Validation of FAA STC SR02024SE: D119-646 Replacement Skidtube Ins. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | STC SR02024SE dated 11 August 2009 MDL-D119-646 Rev. G ICA-D119-646 Rev. 1 IIN-D119-646 Rev. B or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 06.07.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|----------------------------|---|---|---|---|---------------|
| 10041447 | DASSAULT AIRCRAFT SERVICES (DAS) | | | FAA STC ST02964NY-D | MF50 | Installation of Crane Eldec "SmartStem" Tire Pressure Sensor (TPS) System. Validation of FAA STC N° ST02964NY-D. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Dassault Falcon Jet-Wilmington Mater Drawing List, F5XW-D0011-006 Revision I/R dated January 06, 2012; Instructions for Continued Airworthiness doc. N° F5XW-D0011-155 Revision B dated December 13, 2011; FAA STC N° ST02964NY-D issue dated February 21, 20 (...) | None. | 18.09.2012 |
| 10041722 | DASSAULT AIRCRAFT SERVICES (DAS) | | | FAA STC ST02962NY-D | FALCON 2000, FALCON 2000EX | Validation of FAA STC Number ST02962NY-D. Installation of Rockwell Collins, Inc. ADS-B Out System. Description as per Associated Technical Documentation | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | As per Master Document List ref. F2XW-D0006-005 Revision B, dated 19/10/2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None | 11.10.2012 |
| 10041723 | DASSAULT AIRCRAFT SERVICES (DAS) | | | FAA STC ST02963NY-D | MYSTERE-FALCON 50 | Validation of FAA STC Number ST02963NY-D. Installation of Rockwell Collins, Inc. ADS-B Out System. Description as per Associated Technical Documentation. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | As per Master Document List ref. F5XW-D0008-005 Revision A, dated 02/12/2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None | 11.10.2012 |
| 10015765 | DASSAULT FALCON JET | EASA.IM.A.S.01803 | | SA8279SW-D | FALCON 900EX | Installation of the Auxiliary Battery System | | Drawing List Report No. F9EX-01008 Rev. G dated 12-21-2000 or later EASA approved Revisions. FAA approved Airplane Flight Manual Supplement 36500-1, Rev. B dated 2-4-00 or later EASA approved Revisions. Maintenance Manual Supplement 36520& 36606. | | 08.02.2006 |
| 10015766 | DASSAULT FALCON JET | EASA.IM.A.S.01805 | | FAA STC SA8303SW-D | FALCON 900EX | Installation of the Water System Modifications | | Drawing List Report No. F9EX-01017 Rev. A dated 11-04-97 or later EASA approved Revisions | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.02.2006 |
| 10015767 | DASSAULT FALCON JET | EASA.IM.A.S.01806 | | FAA STC SA8270SW-D | FALCON 900EX | Installation of the Miscellaneous Structures (General / Mechanical) | | Drawing List Report No. F9EX-01020 Rev. A dated 01-10-90 or later EASA approved Revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.02.2006 |
| 10015768 | DASSAULT FALCON JET | EASA.IM.A.S.01807 | | FAA STC SA8287SW-D | FALCON 900EX | Installation of the Equipment Rack | | Drawing List Report No. F9EX-01018 Rev. A dated 02-19-97 or later EASA approved Revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.02.2006 |
| 10015778 | DASSAULT FALCON JET | EASA.IM.A.S.01823 | | SA5611SW | MYSTERE FALCON 200 | Installation of the Collins Dual EFIS-85EFIS-85 (Standard II) Instrument & Flight Director System of Collins Dual EFIS-86 Instrument & Flight Director System | | Data package containing the following documents, and additional supporting data: 1. Collins Dual EFIS Instrument and Flight Director System Drawing List, Falcon Jet Report F20-01010, Rev. E.2. FAA STC SA5611SW.3. Airplane Flight Manual Suppl (...) | 1. EASA Approved Airplane Flight Manual Supplement for Collins Dual EFIS-85 (Standard II) Instrument and Flight Director System, Falcon Jet Report 35402-1, Rev. B, or later EASA approved revision is a required part of this STC2. EASA Approved Airpla (...) | 02.08.2006 |
| 10016016 | DASSAULT FALCON JET | EASA.IM.A.S.02182 | | ST10461SC-D | FALCON 900 | Mid Cabin Door Installation, Enflite | | AFMS 37700-1 SMM 37681 | This STC is Limited to F9EX-01056-501 Installations in Model Falcon 900EX. | 27.10.2006 |
| 10016017 | DASSAULT FALCON JET | EASA.IM.A.S.02183 | | ST10510SC-D | FALCON 900 | Aircraft Compliance for Falcon 900EX S/N 165. | | AFMS 37159-F9EX-D165SMM F900EX-SN165 | Limited to S/N 165. No duplicate installation permitted. | 27.10.2006 |
| 10016018 | DASSAULT FALCON JET | EASA.IM.A.S.02184 | | ST10253SC-D | FALCON 900EX | Oxygen System Installation. | | SMM 37140 and SMM 37347 | | 27.10.2006 |
| 10016025 | DASSAULT FALCON JET | EASA.IM.A.S.02193 | | ST09296SC-D | FALCON 2000 | FAA STC ST09296SC-D: "Service and Indirect Lighting System" | | see STC ST09296SC-D: -Drawing List Report F2M-01045 Rev. B dtd 31. Aug.00 or later approved revision | | 18.12.2006 |
| 10016026 | DASSAULT FALCON JET | EASA.IM.A.S.02194 | | SA8267SW-D | FALCON 2000 | FAA STC SA8267SW-D: "Auxiliary Battery System" | | see STC SA8267SW-D: -Drawing List Report F2M-01038 Rev. Htd 5. Jun.01 or later approved revision -AFM Supplement, Report 36482-1, dtd, 9 Aug 96 or later approved revision is required with F2M-01038-501 and -503 installation -AFM Supplement, Repo (...) | | 20.12.2006 |
| 10016027 | DASSAULT FALCON JET | EASA.IM.A.S.02195 | | SA8227SW-D | FALCON 2000 | FAA STC SA8227SW-D: Automatic Electrical Load Shed System | | see STC SA8227SW-D: -Drawing List Report F2M-01004 Rev. Gtd 29. Nov.00 or later approved revision -AFM Supplement, Report 3672-1, Rev. C dtd, 9 Oct 01 or later approved revision | -Systems or components connected to the shed system must have a separate approval different from STC SA8227SW-D -Shedded components are non essential equipment | 18.12.2006 |
| 10016468 | DASSAULT FALCON JET | EASA.IM.A.S.02814 | | FAA STC ST09733SC-D | FALCON 2000 | Installation of Cabin Master Switch. Validation of FAA STC ST09733SC-D. | | Dassault Falcon Jet Drawing List Report No. F2M-01055, Revision B dated 25th May 2002. AFM Supplement Report No. 37023-1, Revision A dated 25th July 2002 is required when F2M-01055-501 and -503 installation is required. AFM Supplement Report No. 37023- (...) | See those of original FAA STC ST09733SC-D Revision 2 dated 04th October 2002. AFM Supplement Report No. 37023-1, Revision A dated 25th July 2002 for F2M-01055-501 and -503 installations. AFM Supplement Report No. 37023-2, Revision NC dated 04th Octobe (...) | 20.05.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|---------------------|----------------|----------------------|-------------------------------------|---|--|---|--|---------------|
| 10016469 | DASSAULT FALCON JET | EASA.IM.A.S.02815 | | FAA STC SA8228W-D | FALCON 2000 | Installation of the Airborne Flight Information System.Validation of FAA STC SA8228W-D. | | Dassault Falcon Jet Drawing List Report No. F2M-1005, Revision B dated 20th May 1999 or later revisions approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | See those of original FAA STC SA8228SW-D Revision 2 dated 21st May 1999 or later revisions approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | 20.05.2008 |
| 10016586 | DASSAULT FALCON JET | EASA.IM.A.S.02991 | | FAA STC ST09313SC-D | FALCON 900EX | EASA Validation of FAA STC ST09313SC-D Revision 1, 01st October 2002Installation of Standby Horizon System Modifications in accordance with Drawing List Report No. F9EX-01035 Revision B dated 25 September 2002 or later revisions approved by EASA i.a (...) | | EASA Approved Airplane Flight Manual Supplement, Report No.37037-1, Revision Original dated 01st October 2002 for the F9EX-01035-503 installation.In addition see those of original FAA STC ST09313SC-D Revision 1 as amended on 01st October 2002 or lat (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 02.04.2009 |
| 10016587 | DASSAULT FALCON JET | EASA.IM.A.S.02992 | | FAA STC ST09311SC-D | FALCON 900EX | EASA Validation of FAA STC ST09311SC-D-Installation of the Inertial Reference System Modification in accordance with Drawing List Report No. F9EX-01033 Revision NC dated 04th May 2000 or later revisions approved by EASA i.a.w. EASA ED Decisions 2007/ (...) | | See those of original FAA STC ST09311SC-D. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 20.03.2009 |
| 10016588 | DASSAULT FALCON JET | EASA.IM.A.S.02994 | | FAA STC SA8291SW-D | FALCON 900EX | EASA Validation of FAA STC SA8291SW-D-Installation of the Battery Installations i.a.w. Drawing List Report No. F9EX-010003 Revision F dated 14 September 2000 or later revisions approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004/04/CF (or subsequent revisions to these decisions) | | EASA Approved Airplane Flight Manual Supplement, Report No.36558-1, dated 01st May 1997 and Airplane Maintenance Manual Supplement, Report No. 36575 dated 09th June 1997, for theF9EX-01003-501 installation.EASA Approved Airplane Flight Manual Supple (...) | 1. The F9EX-01003-503 and F9EX-01003-507 installations are limited to aircraft serial number 049.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previous (...) | 18.03.2009 |
| 10016640 | DASSAULT FALCON JET | EASA.IM.A.S.03080 | | FAA STC ST09293SC-D | FALCON 900EX | EASA Validation of FAA STC ST09293SC-D Revision 10, 22nd December 2004Installation of the FMZ-2000 Flight Management System in accordance with Data List Report No. F9EX-01026 Revision G dated 20 December 2004 or later revisions approved byEASA i.a.w (...) | | See these originally FAA Approved:# Airplane Flight Manual Supplement Report No. 36828-1, dated 07 April 1998 for the F9EX-01026-501 installation.# Airplane Flight Manual Supplement Report No. 36828-2, dated 31 January 2000 for the F9EX-01026- (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. Th (...) | 07.04.2009 |
| 10016643 | DASSAULT FALCON JET | EASA.IM.A.S.03084 | | FAA STC ST09715SC-D | FALCON 900EX | - EASA Validation of FAA STC ST09715SC-D- Installation of the Control Wheel Modifications in accordance with Drawing List Report No. F9EX-01032 | | - FAA STC ST09715SC-D Revision 1, 03rd September 2002*- Drawing List Report No. F9EX-01032 Revision C dated 01st August 2002*- EASA Approved Airplane Flight Manual Supplement, Report No. 37039-1, dated 03rd September 2002 for the F9EX-01032-5 (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. Th (...) | 11.05.2009 |
| 10035017 | DASSAULT FALCON JET | | | FAA STC ST10657SC-D | 1. FALCON 2000EX 2. FALCON 900EX | Assist Handle Installation:Configuration -501: Main entry door assist handle installationConfiguration -503: Alt lavatory assist handle installationValidation of FAA STC N-ST10657SC-D | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/or emissions levels of the original product areunchanged and remain applicable to this certificate/approval. | DFJ Certification Data List doc. N° F00-01013 Revision J dated April 7,2011.Instructions for Continued Airworthiness doc. N° ICA-F00-01013-501/503Revision Original, dated January 25, 2011or later revisions of the above listed documents approved by E (...) | none | 19.05.2011 |
| 10028268 | DASSAULT FALCON JET - WILMINGTON | P-EASA.IM.A.S.03109 | | FAA STC ST01795NY-D | F900EX | EASA Validation of FAA STC ST01795NY-D-Installation of Decrane Aircraft Seating Company, Inc. Model 80A42S2three place partial berthing divan installation in accordance withDassault Falcon Jet-Wilmington Master Drawing List, 197900904 RevisionB, date (...) | The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Instructions for Continued Airworthiness (ICA) 197900902, dated 18thApril 2008 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision2004/04/CF (or subsequent revisions of this decision) | This design change is applicable to Dassault Aviation Mystere Falcon 900aircraft which have previously been equipped with Dassault AviationModification M3000 (Falcon 900EX definition) | 14.12.2009 |
| 10015465 | DASSAULT FALCON JET - WILMINGTON CO | EASA.IM.A.S.01217 | | FAA STC ST01156NY-D | MYSTERE FALCON 900 | Validation of FAA STC ST01156NY-D: Installation of a Honeywell Mark V Enhanced Ground Proximity Warning System (EGPW) upgrade iaw Master Drawing List 027670321, Rev. - dtd Jan 26, 2001. | | Associated to:FAA STC ST01156NY-D.-AFM Supplement 027670302, Rev. - FAA-approved Feb. 12, 2001.- MM Supplement (Continued Airworthiness Instructions) 027670315 Rev - dtd Feb. 12, 2002. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 31.05.2005 |
| 10015466 | DASSAULT FALCON JET - WILMINGTON CO | EASA.IM.A.S.01218 | | FAA STC ST01157NY-D | MYSTERE FALCON 900 | Validation of FAA STC No ST01157NY-D: Installation of a Honeywell Electronic Flight Instrument System (EFIS) ED2-820 Software Upgrade iaw Master Drawing List 027670320, Rev. - dtd. Jan. 26, 2001. | | Associated to:FAA STC ST01157NY-D.-AFM supplement 027670324, Rev. - FAA-approved Feb. 12, 2001.- MM Supplement 027670332 Rev., dtd Feb. 12, 2001. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 31.05.2005 |
| 10015467 | DASSAULT FALCON JET - WILMINGTON CO | EASA.IM.A.S.01219 | | FAA STC ST01775NY-D | MYSTERE FALCON 900 | Validation of FAA STC No ST01775-NY-D: Update of Dual Rockwell Collins TCR-94 Transponder for Elementary Surveillance iaw Master Drawing List 126980303, Rev. - dtd. Oct. 15, 2004. | | Associated to:FAA STC ST01775-NY-D.-AFM supplement 126980302, Rev. -FAA-approved Oct. 18, 2004.-MM Supplement (Cont. Airw. Instructions) 126980307, Rev. - Dtd. Oct. 15, 2004. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 31.05.2005 |
| 10027110 | DASSAULT FALCON JET CORP | P-EASA.IM.A.S.02993 | | FAA STC SA8300SW-D | FALCON 900EX | EASA Validation of FAA STC SA8300SW-D-Installation of the Automatic Cabin Electrical Load Shed System in accordance withDrawing List Report No. F9EX-01023, Revision E dated28th September 2000 or later revisions approved by EASA in accordancewith EASA (...) | - The Certification Basis for the original product remains applicable to this certificate/approval- The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval | - Aeroplane Flight Manual Supplement, Report No. 36571-1, RevisionOriginal dated 05th June 1997 is required for -501 aircraft installations as defined on Drawing List F9EX-1023-Aeroplane Flight Manual Supplement, Report No. 36571-2, Revision Adate (...) | See those of original FAA STC SA8300SW-D | 07.09.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|-------------------------|----------------|----------------------|--------------------|--|---|---|---|---------------|
| 10027112 | DASSAULT FALCON JET CORP | P. EASA.IM.A.S.02997 | | FAA STC SA8283SW-D | FALCON 900EX | EASA Validation of FAA STC SA8283SW-D: Installation of the Auxiliary Light Modifications in accordance with Drawing List Report No. F9CX-1011, dated 16th October 1996 or later revisions approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | - The Certification Basis for the original product remains applicable to this certificate. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate | See those of original FAA STC SA8283SW-D as originally issued on 26th October 1996- or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | See those of original FAA STC SA8283SW-D | 07.09.2009 |
| 10016044 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02215 | | ST09737SC-D | FALCON 900EX | Bracket Installation for Mid Cabin Door i.a.w. Drawing List Report " F90-01093 | | None | | 19.12.2006 |
| 10016216 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02464 | | FAA STC SA8312SW-D | FALCON 50 | Installation of the Automatic Cabin Electrical Load Shed System in accordance with Drawing List Report No. F50-01182 Revision A, dated 27th September 1999 or later approved revisions. | | # FAA STC No. SA8312SW-D# FAA Approved Airplane Flight Manual Supplement Report No. 36527-1 dated 18th February 1998 or later approved revision is required for the -501 installation.# FAA Approved Airplane Flight Manual Supplement Report No. (...) | 1. As per FAA STC No. SA8312SW-D2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon (...) | 19.07.2007 |
| 10016217 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02465 | | FAA STC SA8299SW-D | FALCON 50 | Installation of Auxiliary Battery System in accordance with Drawing List Report No. F50-01173, Revision E dated 19th September 2001 or later approved revisions. | | # FAA STC No. SA8299SW-D.# FAA Approved Airplane Flight Manual Supplement Report No. 36567-1 dated 3rd June 1997 or later approved revision is required for the -501 installation of the Drawing List.# FAA Approved Airplane Flight Manual Supp (...) | 1. As per FAA STC No. SA8299SW-D.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect u (...) | 09.07.2007 |
| 10016218 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02466 | | FAA STC SA8222SW-D | FALCON 50 | Installation of the GWS Satellite Data Communication System in accordance with Drawing List Report No. F50-01159, Revision NC, dated 21st July 1993 or later approved revisions. | | FAA STC No. SA8222SW-D. | 1. As per FAA STC No. SA8222SW-D.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect u (...) | 19.07.2007 |
| 10016219 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02467 | | FAA STC SA7300SW-D | FALCON 50 | Installation of the Service and Indirect Lighting System in accordance with Drawing List Report No. F50-01145, Revision L, dated 25 October 2000 or later approved revisions. | | FAA STC No. SA7300SW-D. | 1. As per FAA STC No. SA7300SW-D.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect u (...) | 17.07.2007 |
| 10016220 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02468 | | FAA STC ST09316SC-D | FALCON 50 | Installation of the Standby Instrument System in accordance with Drawing List Report No. F50-01172, Revision B, dated 12th October 2000 or later approved revisions. | | # FAA STC No. ST09316SC-D# FAA Approved Airplane Flight Manual Supplement Report No. 36606, Revision A, dated 4th May 2000 or later approved revision is required when -501 or -503 installations are used. | 1. FAA STC No. ST09316SC-D2. Limited to Falcon 50EX version aircraft.3. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect u (...) | 19.07.2007 |
| 10016221 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02469 | | FAA STC SA7260SW-D | FALCON 50 | Installation of the Auxiliary Equipment Rack in accordance with Drawing List Report No. F50-01116, Revision N, dated 2nd June 1993 or later approved revisions. | | FAA STC No. SA7260SW-D | 1. As per FAA STC No. SA7260SW-D2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon (...) | 19.07.2007 |
| 10016222 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02470 | | FAA STC SA7069SW-D | FALCON 50 | Installation of the Centralized Water System in accordance with Drawing List Report No. F50-01095-2, Revision A, dated 6th September 1999 or later approved revisions. | | # FAA STC No. SA7069SW-D.# Airplane Maintenance Manual Supplement Report 35699 Revision A or later approved revisions is applicable to those aircraft modified i.a.w. with this STC and having the heated belly drain modification installed as per drawing F50-41813 | 1. As per FAA STC No. SA7069SW-D.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect u (...) | 19.07.2007 |
| 10016524 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02897 | | FAA STC SA7049SW-D | MYSTERE FALCON 900 | Installation of Electrical Feed Thru in Pressure Vessel. | | FAA STC: SA7049SW-D, Revision 10, dated 05 Aug 1994. Drawing List No.: F90-01048, Revision N, dated 19 July 1994. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 23.06.2008 |
| 10016525 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02898 | | FAA STC SA7056SW-D | MYSTERE FALCON 900 | Installation of Automatic Cabin Electrical Load Shed System. | | FAA STC: SA7056SW-D, Revision 6, dated 17 Dec 2000 Drawing List No.: F90-01012, Revision K, dated 12 Dec 2000. For 501 Installation -AFM Supplement No.: 35888-1, Revision B, dated 06 Jan 1988. For 503 or 505 Installation -AFM Supplement No.: 35888-2, Revision A, dated 17 Dec 2000. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 23.06.2008 |
| 10016526 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02899 | REV 1 | FAA STC SA7060SW-D | MYSTERE FALCON 900 | Installation of Basic Interior Furnishings. | | FAA STC: SA7060SW-D, Revision 68, dated 07 Dec 1996 Drawing List No.: F90-01007, Revision F, dated 06 Dec 1996 Maintenance of Pocket Door Assembly: 36343 dated 28 Feb 1994. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 23.06.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|-------------------|----------------|----------------------|--------------------|--|---------------------|--|--|---------------|
| 10016527 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02900 | REV 1 | FAA STC SA7062SW-D | MYSTERE FALCON 900 | Installation of Basic Oxygen System. | | FAA STC: SA7062SW-D, Revision 4, dated 07 Dec 1996 Drawing List No.: F90-01009, Revision R, dated 04 Dec 1996. AFM Supplement No.: 36103-1, dated 20 Jan 1989. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 23.06.2008 |
| 10016528 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02901 | REV 1 | FAA STC SA7085SW-D | MYSTERE FALCON 900 | Installation of Avionics Rack. | | FAA STC: SA7085SW-D, Revision 13, dated 19 Oct 1994 Drawing List No.: F90-01020, Revision AC, dated 21 July 1994. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 23.06.2008 |
| 10016529 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02902 | REV 1 | FAA STC SA7123SW-D | MYSTERE FALCON 900 | Installation of APU Fault Panel Installation. | | FAA STC: SA7123SW-D, Revision 1, dated 20 Sep 1991. Drawing List No.: F90-01024, Revision C, dated 12 Aug 1991. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 23.06.2008 |
| 10016530 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02903 | REV 1 | FAA STC SA7239SW-D | MYSTERE FALCON 900 | Installation of Emergency Power Supply System. | | FAA STC: SA7239SW-D, Revision 3, dated 08 Nov 2000. Drawing List No.: F90-01036, Revision E, dated 03 Nov 2000. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 23.06.2008 |
| 10016531 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02904 | REV 1 | FAA STC SA7227SW-D | MYSTERE FALCON 900 | Installation of Air Distribution System Modification. | | FAA STC: SA7227SW-D, Revision 4, dated 09 Sep 1991. Drawing List No.: F90-01031, Revision T, dated 26 Jun 1991. AFM Supplement No.: 36098-1, dated 20 Jan 1989. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 23.06.2008 |
| 10016532 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02905 | REV 1 | FAA STC ST09308SC-D | MYSTERE FALCON 900 | Installation of FMZ-2000 Flight Management System. | | FAA STC: ST09308SC-D, Revision 2, dated 25 Aug 2003. Drawing List No.: F90-01082, Revision C, dated 15 Jul 2003. For 501 Installation -AFM Supplement No.: 36729-1, dated 13 Dec 1999. For 503 Installation -AFM Supplement No.: 36729-2, dated 16 Oct 2001. Fo (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 23.06.2008 |
| 10016533 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02906 | REV 1 | FAA STC ST09309SC-D | MYSTERE FALCON 900 | Installation of Universal Bracket Attachments. | | FAA STC: ST09309SC-D, Revision 0, dated 26 Jan 2000. Drawing List No.: F90-01081, Revision A, dated 20 Jan 2000. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 23.06.2008 |
| 10016585 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02990 | | FAA STC SA8271SW-D | FALCON 900EX | EASA Validation of FAA STC SA8271SW-D Installation of the APU Fault Panel Installation i.a.w. Drawing List Report No. F9EX-01007 dated 08th October 1996 or later revisions approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004/04/CF (or subsequent revisions to these decisions) | | See those of original FAA STC SA8271SW-D or later revisions of these documents approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004/04/CF (or subsequent revisions to these decisions). | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. (...) | 27.03.2009 |
| 10016589 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.02995 | | FAA STC ST09312SC-D | FALCON 900EX | EASA Validation of FAA STC ST09312SC-D Installation of the Engine Instrument Display System Modification in accordance with Drawing List Report No. F9EX-1034 Revision NC dated 04th May 2000 or later revisions approved by EASA i.a.w. EASA ED Decisions 2 (...) | | See those of original FAA STC ST09312SC-D or later revisions of these documents approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004/04/CF (or subsequent revisions to these decisions). | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 24.03.2009 |
| 10016641 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.03082 | | FAA STC ST09301SC-D | FALCON 900EX | EASA Validation of FAA STC ST09301SC-D Installation of the Service and Indirect Lighting System in accordance with Drawing List report No. F9EX-01027, Revision B dated 28 September 2000 or later revisions approved by EASA i.a.w. EASA ED Decisions 2007 (...) | | See those of original FAA STC ST09301SC-D, originally issued on 28th January 1999 and amended as per Revision 1 on 30 September 2000 or later revisions of these documents approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004/04/CF (or subsequent revisions to these decisions). | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 31.03.2009 |
| 10016642 | DASSAULT FALCON JET CORP. | EASA.IM.A.S.03083 | | FAA STC ST09738SC-D | FALCON 900EX | EASA Validation of FAA STC ST09738SC-D Installation of the HF Radio Master System in accordance with Drawing List Report No. F9EX-01044 Revision NC dated 16 September 2002 or later revisions approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004 (...) | | See those of original FAA STC ST09738SC-D originally issued on 1st October 2002 and EASA Approved Airplane Flight Manual Supplement, Report No. 37035-1, dated 01st October 2002 for the F9EX-01044-501 installation or later revisions of these documents ap (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 31.03.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|---------------------|----------------|------------------------|--------------------------|---|---|---|--|---------------|
| 10016215 | DASSAULT FALCON JET CORPORATION | EASA.IM.A.S.02463 | REV. 1 | FAA STC ST09304SC-D | MYSTERE FALCON 50 (MF50) | Installation of the Inertial Reference System Modifications (installation of Securoplane XL-245 batteries in lieu of Salt batteries to provide backup power) in accordance with Drawing List report No.F50-01188 Revision NC dated 23rd July 1999 or later approved revisions. | | FAA STC ST9304SC-D.Airplane Maintenance Manual Supplement Report No. 36606 (Securoplane XL245 Battery) dated 30th June 1999 or later approved revisions. | As per FAA STC ST09304SC-D | 04.05.2012 |
| 10016224 | DASSAULT FALCON JET CORPORATION | EASA.IM.A.S.02472 | | FAA STC ST10423SC-D | FALCON 900EX | Installation of the Service and Indirect Lighting System (EMTEQ) in accordance with Data List Report No. F9EX-01048 Revision A dated 07th October 2005 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decis (...) | | FAA STC ST10423SC-D Maintenance Manual Supplement, Report 37498 Revision B dated 26th September 2005, or later revisions approved by EASA in accordance with EASA ED Decision 2004/04/CF (or i.a.w. subsequent revisions of this decision) | 1. As per FAA STC ST10423SC-D.2. Applicable to Falcon900EX EASy aircraft version.3. F900EX EASy designation does not correspond to a new MF900 model designation but is commercial designation for airplanes on which the specific Dassault modl (...) | 26.09.2007 |
| 10027395 | DASSAULT FALCON JET CORPORATION | P-EASA.IM.A.S.03264 | | FAA STC SA8279SW-D | MF900, F900EX | EASA Validation of FAA STC SA8279SW-D: Installation of the Auxiliary Battery System in accordance with Drawing List Report No. F9EX-01008, Revision K dated 26th September 2002 or later revisions approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | The Certification Basis for the original product remains applicable to this certificate/ approval. Define the additional airworthiness requirements SCs and accepted ESFs. The certificated noise and/ or emissions levels of the original product are unchanged (...) | Drawing List Report No. F9EX-01008, Revision K dated 26th September 2002. Aeroplane Flight Manual Supplement, Report No. 36500-1, Revision B dated 04th February 2000, or later revisions of the above listed documents approved by EASA in accordance with E (...) | This design change is also applicable on the Falcon 900C designation as defined on EASA TCDS.A.062 | 28.09.2009 |
| 10028777 | DASSAULT FALCON JET CORPORATION | EASA.IM.A.S.02998 | | FAA STC SA8275SW-D | FALCON 900EX | | | | | 16.07.2009 |
| 10029434 | DASSAULT FALCON JET CORPORATION | P-EASA.IM.A.S.02996 | | FAA STC SA8285SW-D | FALCON 900EX | EASA Validation of FAA STC SA8285SW-D Revision 1: Installation of the Security System (Structural Provisions) in accordance with Drawing List Report No. F9EX-01021, Revision B dated 13th June 1997 or later revisions approved by EASA in accordance with EA (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | See those of original FAA STC 8285SW-D Revision 1 as amended by the FAA on 14th June 1997 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None. | 24.03.2010 |
| 10035016 | DASSAULT FALCON JET CORPORATION | | | FAA STC N° ST10262SC-D | F2000EX | Oxygen system installation. Installation of added portion of the continuous flow, dropout type oxygen system for the cabin and lavatory. This system connects to the existing TC aircraft oxygen system. Validation of FAA STC N° ST10262SC-D, Revision 2 Issue date April 14, 2011 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | DFJ Certification Data List doc. N° F2EX-01006 Revision K dated March 31, 2011. Instructions for Continued Airworthiness doc. N° ICA-F2EX-01006-503 Revision Original, dated January 25, 2011, or later revisions of the above listed documents approved by E (...) | As per FAA STC N° ST10262SC-D, Revision 2 Issue date April 14, 2011. | 19.05.2011 |
| 10035018 | DASSAULT FALCON JET CORPORATION | | | FAA STC N° ST10584SC-D | F2000EX & F900EX | Console table installation. Installation of a Console Table in the main cabin area of the aircraft side ledge. This table design will typically be located between two single seats and is not designed to be installed in front of the emergency escape hatch. Validation of FAA STC N° ST10584SC-D | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | DFJ Certification Data List doc. N° F00-01008 Revision K dated April 2, 2011. Instructions for Continued Airworthiness doc. N° ICA-F00-01008-501 Revision Original, dated January 25, 2011, or later revisions of the above listed documents approved by EASA (...) | None | 19.05.2011 |
| 10035461 | DASSAULT FALCON JET CORPORATION | | | FAA STC N° ST10583SC-D | 1. F2000EX 2. F900EX | Galley pocket door installation. Validation of FAA STC N° ST10583SC-D | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | DFJ Certification Data List doc. N° F00-01005 Revision AR dated April 03, 2011. Instructions for Continued Airworthiness doc. N° ICA-F00-01005-501 Revision Original, dated January 25, 2011. Airplane Flight Manual Supplement doc. N° 37887-1 dated July 16, 2 (...) | Approval is limited to the "501" configuration. | 24.06.2011 |
| 10037659 | DASSAULT FALCON JET CORPORATION | | | FAA STC ST10890SC-D | F2000EX | Alt RH Divan installation. Validation of FAA STC N° ST10890SC-D | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Certification Review Item CRI D-2112-001 closed at issue 3 - Special Conditions "Single (...) | DFJ Certification Data List doc. N° F2EX-01038 Revision AJ dated October 31, 2011. Instructions for Continued Airworthiness doc. N° ICA-F2EX-01038-501 Revision Original, dated January 21, 2011. Airplane Flight Manual Supplement doc. N° 40709-1 dated April (...) | As per FAA STC N° ST10890SC-D | 14.12.2011 |
| 10037660 | DASSAULT FALCON JET CORPORATION | | | FAA STC ST10891SC-D | F2000EX | Avionics and interiors installation. Validation of FAA STC N° ST10891SC-D | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI D-GENE-03 complied with through compliance with FAR 25 §25.856(a) The requirements (...) | DFJ Certification Data List doc. N° F2EX-01039 Revision C dated December 07, 2011. Supplemental Maintenance Manual report N° SMM F2000EX-S/N216, Revision Original, dated January 21, 2011. Airplane Flight Manual Supplement doc. N° 37159-F2EX-D514 dated April (...) | As per FAA STC N° ST10891SC-D Restricted to F2000EX S/N 216 | 14.12.2011 |
| 10043077 | DASSAULT FALCON JET CORPORATION | EASA.IM.A.S.03081 | | FAA STC ST09736SC-D | FALCON 900EX | EASA Validation of FAA STC ST09736SC-D: Installation of a Cabin Master Switch System i.a.w. Drawing List Report No. F9EX-1041 | | - FAA STC ST09736SC-D original 06th September 2002- EASA Approved Drawing List Report No. F9EX-1041 Revision B dated 29th August 2002- EASA Approved Aeroplane Flight Manual Supplement Report No. 37026-1, Revision NC dated 06th September 2002 or later (...) | none | 20.05.2009 |
| 10013990 | DASSAULT FALCON SERVICE | EASA.A.S.01022 | | | MYSTERE FALCON 20-5 | Installation of Collins TCAS-4000 and of 2 ATC TDR 94D capable of elementary surveillance. | | - Change Approval Sheet DFS 533 issue original dated 7 April 2005. - AFM Supplement DFS No 31 issue original dated 7 January 2005. | | 28.04.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|----------------|----------------|----------------------|---------------------|--|---------------------|---|---|---------------|
| 10014018 | DASSAULT FALCON SERVICE | EASA.A.S.01065 | | | MYSTERE FALCON 50 | Installation of a Honeywell Mark V EGPWS soft 217-217 with terrain display in the Collins EFIS 86-C14 | | -Change Approval Sheet DFS 5601 issue A dated January 19th 2005-AFM Supplement DFS No 49 issue original dated December 29th 2004 | 1. Applicable to Mystère Falcon 50 equipped with Collins EFIS 86-C14 only2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with otheraircraft/en (...) | 03.02.2005 |
| 10014030 | DASSAULT FALCON SERVICE | EASA.A.S.01096 | | | MYSTERE FALCON 50 | Installation of a Honeywell Mark VII EGPWS soft 217-217 with terrain display in the MFD | | -Change Approval Sheet DFS 5603 issue original dated February 8th 2005-AFM Supplement DFS No.50 issue original dated February 8th 2005-if the aircraft has steep approach capability AFM Supplement DFS No. 51 issue original dated February 8th 2005 | This STC is only valid for Falcon 50 equipped with a Universal FMS UNS-1B with GPS-1000, or UNS-1C/1C+/1D/1E/1F/1K/1L and with Collins EFIS-86C(x)/weather radar WXR-350 | 15.02.2005 |
| 10014052 | DASSAULT FALCON SERVICE | EASA.A.S.01134 | | | FALCON 10 | DFS 1261 dtd 13.Oct.2005: Installation of a TAWS Honeywell Mark VIII EGPWS soft 011 | | see Change approval sheet DFS 1261-AFM Supp F10 DFS 29 toDTM 721(A) with DFS 1261 incorp "TAWS Honeywell Mark VIII Soft 011"-AFM Supp F10 DFS 31 to DTM 721(A) with DFS 1261 incorp "EGPWS Steep Approach Function"-Wiring Diagram DFS 10 600 31-56-0 (...) | 1. This STC is approved only for the configuration as defined in Mod Sheet DFS 1261.2. Compatibility with other aircraft/engine configurations shall be determined by the installer | 02.05.2006 |
| 10014071 | DASSAULT FALCON SERVICE | EASA.A.S.01161 | | | FALCON 10 | DFS 1255 dtd. 29.3.06: Installation of two FMS UNS-1L Soft 802.X / B-RNAV & P-RNAV | | see Change approval sheet DFS 1255-AFM Supp F10 DFS 32 toDTM 721(A) "Two FMS UNS-1L Operation / DFS 1255 incorporated"-Wiring Diagram DFS 10 600 34-56-01 & DFS 10 600 34-56-11and associated list of affected technical documents | 1. This STC is approved only for the configuration as defined in Mod Sheet DFS 1255. Compatibility with other aircraft/engine configurations shall be determined by the installer | 02.05.2006 |
| 10014072 | DASSAULT FALCON SERVICE | EASA.A.S.01163 | REV. 1 | | FALCON 10 | DFS 1252 dtd. 9.6.05: EFIS Universal EFI 550/600 | | see Change approval sheet DFS 1252-AFM Supp F10 DFS 30 toDTM 721(A) with DFS 1252 concernant utilisation de l'avion equipé de EFI UNIVERSAL 550/600"-Wiring Diagram DFS 10 600 24-61-01, 34-21-10, 34-21-20 and associated list of affected technical documents | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and #Associated Technical Documentation#. Compatibility with other aircraft/engine configurations shall be determined by theinstaller. | 11.01.2007 |
| 10014099 | DASSAULT FALCON SERVICE | EASA.A.S.01200 | | | FALCON 10 | Installation of Collins ATC with flight ID and Gables Control Panel | | -Change Approval Sheet DFS 1260 issue original dated 31 March 2005. | STC applicable only to Falcon 10 aircraft equipped with:- 2Collins ATC TDR-94D p/n 622-9210-001/002/003 or 004.- and ATC control box Gables G7 130-XX. | 11.04.2005 |
| 10014117 | DASSAULT FALCON SERVICE | EASA.A.S.01224 | | | FALCON 10 | DFS 1251 dtd. 2.6.05: Installation of a Multi functional display MFD 640 | | see Change approval sheet DFS 1251-Wiring Diagram DFS 10 600 34-41-21and associated list of affected technical documents | 1. This STC is approved only for the configuration as defined in Mod Sheet DFS 1251. Compatibility with other aircraft/engine configurations shall be determined by the installer | 02.05.2006 |
| 10014118 | DASSAULT FALCON SERVICE | EASA.A.S.01225 | | | MYSTERE FALCON 900 | EGPWS Software Evolution (222-222) | | Change Approval Sheet DFS 9472 CAS issue Original dated 13-September 05Airplane Flight Manual DTM 20103 DFS Supplement No 21 | Applicable to MF900 equipped with EGPWS Mark V soft 212-212installed with DFS 9259 STC | 12.12.2005 |
| 10014119 | DASSAULT FALCON SERVICE | EASA.A.S.01226 | | | MYSTERE FALCON 900 | Incorporation of RAAS function on EGPWS soft 222-222 | | Change Approval Sheet DFS 9470 CAS issue Original dated April 14, 2006Airplane Flight Manual DTM 20103 DFS Supplement n°22 dated June 14, 2006 | Applicable to MF900 equipped with EGPWS Mark V soft 222-222installed with DFS 9472 STC | 06.10.2006 |
| 10014137 | DASSAULT FALCON SERVICE | EASA.A.S.01248 | | | FALCON 10 | DFS 1250 dtd. 22.3.06: Weather Radar Primus 660 | | see Change approval sheet DFS 1250-Wiring Diagram DFS10 60034-41-21-XXand associated list of affected technical documents | 1. This STC is approved only for the configuration as defined in Mod Sheet DFS 1250. Compatibility with other aircraft/engine configurations shall be determined by the installer | 02.05.2006 |
| 10014141 | DASSAULT FALCON SERVICE | EASA.A.S.01254 | | | MYSTERE FALCON 20-5 | Installation of one FMS Universal UNS-1L SOFT 802.X in accordance with the Change Approval Sheet -Original- or later EASA approved revisions. | | -Change Approval Sheet DFS 529, dated May 03, 2005-AFM Supplement No 32,Temporary AFMS dated May 12th 2005 or later EASA approved revisions | Operation of the Flight Management System in the Approach Mode is not allowed | 19.05.2005 |
| 10014152 | DASSAULT FALCON SERVICE | EASA.A.S.01273 | | | FALCON 2000 | DFSD066 dtd. 19. October 05: Airview Communication and Security System | | see Change approval sheet DFSD066:DFS AFM Supplement F20003 to DTM 537and associated list of affected technical documents | 1. This STC is limited to Falcon 2000 S/N 0012. This STC isapproved only for the configuration as defined in Mod SheetDFSD066 | 08.11.2005 |
| 10014156 | DASSAULT FALCON SERVICE | EASA.A.S.01280 | | | MYSTERE FALCON 200 | AFM supplement associated to Dassault Aviation Mod M1632: "Installation d'un hublot inférieur avec volet de protection". | | Subject AFM supplement is associated to Dassault Aviation Mod M1632: "Installation d'un hublot inférieur avec volet de protection", applicable to F20 GF SN 397 only | Subject AFM supplement is associated to Dassault Aviation Mod M1632: "Installation d'un hublot inférieur avec volet de protection", applicable to F20 GF SN 397 only | 21.06.2005 |
| 10014170 | DASSAULT FALCON SERVICE | EASA.A.S.01303 | | | FALCON 10 | Antenna installation | | Associated to: see Change approval sheet DFS 1263 associated list of affected technical documents | 1. This STC is approved only for the configuration as defined in Mod Sheet DFS 1263.2. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 10.08.2005 |
| 10014172 | DASSAULT FALCON SERVICE | EASA.A.S.01305 | | | MYSTERE FALCON 50 | Introduction of Pressure Seals for wire runs | | -see Change approval sheet DFS 5610 and associated list of affected technical documents | This STC is approved only for the configuration as defined in Mod Sheet DFS 5610. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.09.2005 |
| 10014217 | DASSAULT FALCON SERVICE | EASA.A.S.01378 | | | MYSTERE FALCON 900 | Call out evolution with EGPWS Honeywell Mark V 212-212. ThisSTC supersedes the STC F-C163-SF-2107 EASA approved on March 9, 2004 with the reference 2004-2073. | | Change Approval Sheet DFS 9447 CAS revision A dated 9 June 2005.AFM supplement DFS n° 17 revision A dated June 9, 2005. | Applicable to all MF900 aircraft with modification DFS 9259 | 04.07.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|----------------|----------------|----------------------|---|--|---|---|--|---------------|
| 10014243 | DASSAULT FALCON SERVICE | EASA.A.S.01452 | | | MYSTERE FALCON 20 | Deviation to M GE 2 - Pressure seals FR17-18 | | see Change approval sheet DFS 2561 and associated list of affected technical documents | 1. This STC is applicable to Falcon 20S/N 103 only. This STC is to be applied simultaneously with Dassault Aviation Mod M GE23. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.09.2005 |
| 10014255 | DASSAULT FALCON SERVICE | EASA.A.S.01481 | | | FAN JET MYSTERE FALCON 20 | DFS 2545 dtd. 12. October 05: Camera and Video Recorder installation | | see Change approval sheet DFS 2545 and associated list of affected technical documents | This STC approves the installation only, system activation is not considered. | 08.11.2005 |
| 10014263 | DASSAULT FALCON SERVICE | EASA.A.S.01493 | | | | Installation of two V/UHF AN/ARC210 | | Associated to: see Change approval sheet DFS 2543 associated list of affected technical documents | 1. This STC is applicable to Falcon 20 SN 076 only. 2. System activation is not considered. 3. This STC is approved only for the configuration as defined in Mod Sheet DFS 25434. Compatibility with other aircraft/engine configurations shall be determined. | 09.08.2005 |
| 10014267 | DASSAULT FALCON SERVICE | EASA.A.S.01496 | | | MYSTERE FALCON 20 | Antenna relocations, skin repairs | | see Change approval sheet DFS 2558 and associated list of affected technical documents | 1. This STC is only approved for the configuration as defined in Mod Sheet DFS 2558 Issue Original | 14.09.2005 |
| 10014270 | DASSAULT FALCON SERVICE | EASA.A.S.01500 | | | FALCON 10 | GPS Antenna installation | | Associated to: see Change approval sheet DFS 2563 and associated list of affected technical documents | 1. Applicable to Falcon 20 S/N 103 only. | 16.08.2005 |
| 10014274 | DASSAULT FALCON SERVICE | EASA.A.S.01510 | | | FALCON 20 | Time Base Installation. | | Associated to: see Change approval sheet DFS 2548 and associated list of affected technical documents | 1. This STC is applicable to Falcon 20 SN 076 only. 2. System activation is not considered. 3. This STC is approved only for the configuration as defined in Mod Sheet DFS 2548 4. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.08.2005 |
| 10014276 | DASSAULT FALCON SERVICE | EASA.A.S.01512 | | | FALCON 10 | Frame 0 Pressure Seal through-fitting modification (enlarge pressure seals) | | Associated to: see change approval sheet DFS 1272 | This STC is approved only for the configuration as defined in Mod Sheet DFS 1272 Issue Original. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 24.08.2005 |
| 10014284 | DASSAULT FALCON SERVICE | EASA.A.S.01521 | | | MYSTERE FALCON 20 | Power Utility | | see Change approval sheet DFS 2549 associated list of affected technical documents | System activation is not considered | 28.09.2005 |
| 10014294 | DASSAULT FALCON SERVICE | EASA.A.S.01536 | | | MYSTERE FALCON 20 | DFS 2562 dtd 21. Dec 05: Master Mission Function | | see Change approval sheet DFS 2562: DFS AFM Supplement F2049 and associated list of affected technical documents | 1. This STC is approved only for the configuration as defined in Mod Sheet DFS 2562. 2. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.01.2006 |
| 10014332 | DASSAULT FALCON SERVICE | EASA.A.S.01586 | | | FALCON 10 | Update ATC Transponder for elementary surveillance DFS 1253 issue A | | See Change approval sheet DFS 1253 associated list of affected technical documents | This STC is approved only for the configuration as defined in Mod Sheet DFS 1253. | 10.10.2005 |
| 10014334 | DASSAULT FALCON SERVICE | EASA.A.S.01589 | | | FALCON 10 | DFS 1226 Revision D dated 9 June 2005: Installation of Collins TCAS 4000 with one or two TDR-94D | | 1. See Change approval sheet DFS 1226 associated list of affected technical documents. 2. AFM Supplement F10 DFS No 34 dtd. 09.06.2005 | This STC is approved only for the configuration as defined in Mod Sheet DFS 1226. | 10.10.2005 |
| 10014383 | DASSAULT FALCON SERVICE | EASA.A.S.01667 | | | MYSTERE FALCON 2000 MYSTERE FALCON 2000EX | DFS D078 dtd. 5.4.2006: EGPWS Software 222-222 | | see Change approval sheet DFS D078 and associated list of affected technical documents | 1. Activation of Runway Awareness Advisory System (RAAS) is not approved through this STC. 2. This STC is approved only for the configuration as defined in Mod Sheet DFS D0782. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 07.07.2006 |
| 10014436 | DASSAULT FALCON SERVICE | EASA.A.S.02485 | | | | DFS 2568 dtd. 21. Dec 05: DFS AFM Supplement F20 51 to DTM 589/DTM590/DTM591 | | see Change approval sheet DFS 2568: DFS AFM Supplement F20 51 to DTM 589/DTM590/DTM591 and associated list of affected technical documents | 1. This STC is approved only for the configuration as defined in Mod Sheet DFS 2568. 2. Compatibility with other aircraft/engine configurations shall be determined by the installer. 3. This STC is approved only for the product configuration as defined (...) | 26.01.2006 |
| 10014505 | DASSAULT FALCON SERVICE | EASA.A.S.02575 | REV. 1 | | MF900 & F900EX | 220 VAC ELECTRICAL DISTRIBUTION ON AIRCRAFT STC EASA.A.S 02575 revision for applicability extension and addition of version 2. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate approval: §25.669 (a) (1)&(4) at JAR 25 Change 14. The requirements for environmental protection a (...) | DFS Change Approval Sheet DFS 59511-CAS Issue A dated 12.10.2010 DFS Modification Sheet DFS 9511, Issue A dated 07.10.2010. DFS Airplane Flight Manual Supplement (Falcon 900) N°24 Rev. 2 dated 07.10.2010 or later revisions of the above listed documents approved by EASA | None | 13.10.2010 |
| 10014517 | DASSAULT FALCON SERVICE | EASA.A.S.02593 | | | MYSTERE FALCON 50 | DFS 5625A dtd. 4. Mai 2006: Equipment installation with 220V inverter | | see Change approval sheet DFS 5625A and associated list of affected technical documents | | 05.05.2006 |
| 10014529 | DASSAULT FALCON SERVICE | EASA.A.S.02614 | | | FALCON 10 | DFS 1274 dtd 5.5.06: Mode S Transponder Enhanced Surveillance | | see Change approval sheet DFS 1274-AFM Supp F10 DFS 35 to DTM 721(A) with DFS 1274 incorp "Mode S Transponder Enhanced Surveillance"-Wiring Diagram DFS10 600 34-56-01XX, DFS10 600 34-56-11XX (Optional), DFS10 600 34-42-03XX, DFS10 600 34-16-01XX and associated list of affected technical documents | Applicable to Falcon 10 with DFS 1269B applied previously or simultaneously | 05.05.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|----------------|----------------|----------------------|---|---|---------------------|---|--|---------------|
| 10014617 | DASSAULT FALCON SERVICE | EASA.A.S.02747 | | | FAN JET MYSTERE FALCON 20 MYSTERE FALCON 20-5 MYSTERE FALCON 200 | DFS536 Issue A dtd. 29.Sep 06: Honeywell EGPWS Mark V installation with Terrain display in Collins EFIS 86-C14 | | see Master Document List DFS-DE-17-F3-536 Issue A1 dtd.29.Sep 06 or later approved revisions.-AFM Supplement 33 to DTM30528 Rev. Original or later approved revisions-associated list of affected technical documents | Leaflet TGL 12 "Certification Considerations for the Terrain Awareness Warning System: TAWS" applies | 02.10.2006 |
| 10014619 | DASSAULT FALCON SERVICE | EASA.A.S.02751 | | | FAN JET MYSTERE FALCON 200 | DFS2574: Limitation Removal | | see Change approval sheet DFS2574:- AFM Supp FJF 44 issue4 or later approved revision and associated list of affected technical documents | Applicable to Fan Jet Falcon S/N 328 only | 05.09.2006 |
| 10014666 | DASSAULT FALCON SERVICE | EASA.A.S.02823 | | | FALCON 10 | MODE S TRANSPONDER WITH ENHANCED SURVEILLANCE CAPACITY | | See Master Document List DFS-DE-17-F1-1278 Issue B1 or later approved revisions Aircraft Flight Manual Supplement DFS N°36, Rev Original or later approved revisions | 1. This STC is approved only for those Falcon 10 aircraft equipped with two Collins TDR-94D(-007) Mode #S# ATC Transponders and #Flight Ident# source.2. Compatibility with previous approved modifications shall be determined by the installer.3. JAA L (...) | 05.12.2006 |
| 10014678 | DASSAULT FALCON SERVICE | EASA.A.S.02839 | | | MYSTERE FALCON 900 | MODE S TRANSPONDER WITH ENHANCED SURVEILLANCE CAPACITY | | Master Document List DFS-DE-17-F9-9517 Issue B1 or later approved revisions Aircraft Flight Manual Supplement DFS N° 25, Rev Original or later approved revisions | This STC is approved only for those Mystere Falcon 900B aircraft equipped with two Collins TDR-94D(-007) Mode #S# ATC Transponders and #Flight Ident# source. Compatibility with previous approved modifications shall be determined by the installer. JAA (...) | 30.11.2006 |
| 10014682 | DASSAULT FALCON SERVICE | EASA.A.S.02843 | | | FALCON 10 | Extended Function (P-RNAV / RNP 1 Compliance) on Flight Management System (FMS) UNS-1K+ installation | | # DFS Master Document List DFS-DE-17-F1-1279 Issue A1.# Aircraft Flight Manual Supplement DFS F10-037 Revision Original or later approved revisions for those Falcon 10 aircraft on which this STC has been embodied or later approved revisions of these documents. | This STC is applicable to Falcon 10 aircraft equipped with:# Collins Autopilot and Flight Director APS80,# Collins EFIS 85C-14.# FMS Universal UNS-1K+ with software version 802.2 or later FMS Universal UNS-1K+ installation has to be embodied (...) | 29.07.2008 |
| 10014685 | DASSAULT FALCON SERVICE | EASA.A.S.02848 | | | MYSTERE FALCON 50 | MODE S TRANSPONDER WITH ENHANCED SURVEILLANCE CAPACITY | | See Master Document List DFS-DE-17-5628 Issue A1 or later approved revisions Aircraft Flight Manual Supplement DFS N° 52, Rev Original or later approved revisions | This STC is approved only for those Mystere Falcon 50 aircraft equipped with:- Two Collins TDR-94D(-007) Mode #S# ATC Transponders and #Flight Ident# source, and Two Inertial Reference Systems. Compatibility with previous approved modification (...) | 29.11.2006 |
| 10014687 | DASSAULT FALCON SERVICE | EASA.A.S.02850 | | | FAN JET FALCON SERIES C MYSTERE FALCON 20-C5 MYSTERE FALCON 20-D5 MYSTERE FALCON 20-E5 MYSTERE FALCON 20-F5 | Mode S Transponder with enhanced surveillance capacity. | | # Master Document List DFS-DE-17-F3-539 Issue A1 or later approved revisions# Aircraft Flight Manual Supplements: DFS55, for Fan Jet Falcon aircraft, Rev Original or later approved revisions DFS34, for Mystere Falcon 20-(J5, Rev Original or later approved revisions | 1. This STC is approved only for those Fan Jet Falcon and Mystere Falcon 20-(J5 aircraft equipped with two Collins TDR-94D(-007) Mode #S# ATC Transponders and #Flight Ident# source.2. JAA Leaflet TGL 13 #Certification of Mode S Transponder Syst (...) | 18.01.2007 |
| 10014688 | DASSAULT FALCON SERVICE | EASA.A.S.02851 | | | MYSTERE FALCON 900 | CABIN REFURBISHMENT | | See Master Document List DFS-DE-17-F9-9523 Issue A1 or later approved revisions | This STC is approved only for Mystere Falcon 900 Serial Number Aircraft 054. Compatibility with previous approved modifications shall be determined by the installer. | 21.12.2006 |
| 10014714 | DASSAULT FALCON SERVICE | EASA.A.S.02887 | | | FAN JET FALCON SERIES C FAN JET FALCON SERIES D FAN JET FALCON SERIES E FAN JET FALCON SERIES F FAN JET FALCON SERIES G | Incorporation of Runway Awareness and Advisory System (RAAS) Function on EGPWS Honeywell Mark V or Mark VII. | | Compliance Checklist DFS2576-CCL Issue A1 or later approved revisions. Master Document List DFS-DE-17-F2-2576 Issue A1 or later approved revisions. Airplane Flight Manual Supplement DFS N°53 Revision original or later approved revisions for those aircraft (...) | This STC is approved for all Fan Jet Falcon all series and all Mystere Falcon 20-(J5 aircraft with EGPWS Mark V or Mark VII software 224-224 and Terrain Database or later. Aircraft avionics required pre-installation as per DFS Change Approval Sheet DFS2576-CAS Issue A1 or later approved revisions. | 18.07.2007 |
| 10014733 | DASSAULT FALCON SERVICE | EASA.A.S.02915 | | | MYSTERE FALCON 50 | Mode S Transponder with enhanced surveillance capacity. | | Master Document List DFS-DE-17-F5-5630 Issue A1 or later approved revisions Aircraft Flight Manual Supplement DFS N° 56, Rev Original or later approved revisions. | This STC is approved only for those Mystere Falcon 50 aircraft equipped with two Collins TDR-94D(-007) Mode #S# ATC Transponders, #Flight Ident# source and without IRS and with one or two FMSs. Compatibility with previous approved modifications shall (...) | 08.02.2007 |
| 10014820 | DASSAULT FALCON SERVICE | EASA.A.S.03030 | | | | Elementary and enhanced mode S transponder capabilities on existing dual MST-67A mode S transponder installation. | | DFS Master Document List DFS-DE-17-F4-H268 Issue B1 Mystere Falcon 200 AFM Supplement DFS 12 Original Issue Maintenance Manual Supplement.# Supplement 5-20/34 Revision 1# SMM34-95-05A2 and SMM 34-95-15A2 Revision 1 or later approved revisions of these documents. | This STC is limited to all Mystere Falcon 200 aircraft equipped with two Mode S Transponders Honeywell MST#67A with antenna diversity as part of the original transponder installation plus two FMS Universal UNS-1() and two ADC 80 installed. See: Myster (...) | 22.04.2008 |
| 10014847 | DASSAULT FALCON SERVICE | EASA.A.S.03069 | | | MYSTERE FALCON 50 | Incorporation of Runway Awareness and Advisory System (RAAS) function on EGPWS Honeywell MK V. | | # Compliance Checklist DFS 5633-CCL Issue A1 or later approved revisions.# Master Document List DFS-DE-17-F5-5633 Issue A1 or later approved revisions.# Aircraft Flight Manual Supplement DFS N° 54, Rev Original or later approved revisions for (...) | 1. This STC is approved only for those Mystere Falcon 50 aircraft equipped with a Honeywell EGPWS Mk V with Software version 212-212 or later and with Proline Avionic Suite V6 with Terrain Display available; Terrain Database 435 or later is require (...) | 30.05.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|------------------|----------------|----------------------|------------------------------------|--|---|--|--|---------------|
| 10015006 | DASSAULT FALCON SERVICE | EASA.A.S.03306 | | | FALCON 2000 FALCON 2000EX | F2000 / F2000EX Incorporation of Runway Awareness and Advisory System (RAAS) Function on EGPWS Honeywell Mk V. | | 1. DFS Master Document List DFS-DE-17-FD-D100 Issue A1 or later approved revisions.2. DFS Change Approval Sheet DFS-D100-CAS Issue A1 or later approved revisions.3. DFS AFM Supplement DFS-F2000-005 Original Issue or later approved revisions. | 1. As per applicability defined on DFS Change Approval Sheet DFS-D100-CAS Issue A1 or later approved revisions this STC is limited to those Falcon 2000 and Falcon 2000EX previously equipped with:- Honeywell EGPWS Mk V Soft. 212-212 or later w/ (...) | 21.11.2007 |
| 10015071 | DASSAULT FALCON SERVICE | EASA.A.S.03397 | | | MYSTERE FALCON 50 | Air Data Sensor Installation for non essential systems | | DFS Master Document List DFS-DE-17-F5-5657 Issue A1 or later approved revisions. | This STC is limited to all Mystere Falcon 50 aircraft without ARINC 429 air data parameters available. In case the aircraft is RVSM compliant this DFS modification will be applicable only on those aircraft on which Dassault Aviation SB F50-246 have been implemented. | 28.03.2008 |
| 10015108 | DASSAULT FALCON SERVICE | EASA.A.S.03451 | REV. 1 | | MYSTERE FALCON 50 | IFE / Galley Electrical Load Shedding. | | 1. For Falcon 50:- DFS Master Document List DFS-DE-17-F5-5653 Issue A1 or later approved revisions;- Aircraft Flight Manual Supplement F50-058 Revision Original or later approved revisions.2. For Falcon 50EX:- DFS Master Document List (...) | 1. This STC is applicable to:- All Mystere Falcon M50 aircraft and;- Falcon 50EX aircraft variant up to serial number aircraft 333 included, on which Dassault Aviation Mod. M2849 #IFE Load Shedding; electrical capability# has not been embed (...) | 26.06.2008 |
| 10015110 | DASSAULT FALCON SERVICE | EASA.A.S.03453 | | | MYSTERE FALCON 50 | Installation of AC Electrical Generation / Distribution compliant with JAA TGM/25/10 requisites | | DFS Master Document List DFS-DE-17-F5-5651 Issue A1 or later approved revisions. | STC applicable on those Mystere Falcon 50 aircraft on which STC EASA.A.S.03451 (Mod. DFS 5653) #IFE / Galley Electrical Load Shedding# has been implemented previously or simultaneously. | 28.03.2008 |
| 10015181 | DASSAULT FALCON SERVICE | EASA.A.S.03560 | | | FALCON 900EX | Falcon 900C and F900EX: Incorporation of Runway Awareness and Advisory System (RAAS) Function software version 224-224 on EGPWS Honeywell Mk V | | DFS Master Document List DFS-DE-17-F9-9541 Issue A1 or later approved revisions.DFS Change Approval Sheet DFS9541-CAS Issue A1 or later approved revisions.DFS AFM Supplement F900-027 Original Issue (dated 09th June 2008) or later approved revisions. (...) | As per applicability defined on DFS Change Approval Sheet DFS9541-CAS Issue A1 or later approved revisions this STC is limited to those Falcon 900C and Falcon 900EX aircraft previously equipped with:# Honeywell EGPWS Mk V Soft. 212-212 with terra (...) | 10.06.2008 |
| 10015266 | DASSAULT FALCON SERVICE | EASA.A.S.03736 | | | FALCON 900EX MYSTERE FALCON 900 | One or two three-places Sofa installation in the aft part of the Cabin. | | DFS Master Document List DFS-DE-17-F9-9549 Issue A1 or later approved revisions.DFS Change Approval Sheet DFS9549-CAS Issue A1 or later approved revisions. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. (...) | 09.01.2009 |
| 10027351 | DASSAULT FALCON SERVICE | P-EASA.A.S.03939 | | | MYSTERE FALCON 50 | P-RNAV Capability | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Compliance has been shown in accordance with the requisites laid down by JAA Temporary (...) | -DFS Change Approval Sheet DFS 5659-CAS Issue A, dated 21st September 2009.-DFS Aeroplane Flight Manual Supplement Number 48, Issue 2, dated 21st September 2009 | This EASA approval is applicable on those Mystere Falcon 50 aircraft which have previously incorporated the Mod DFS 5598 (approved under EASA.A.S.00615) with:-, Collins Autopilot APS 85,-, EFIS 86 C14 and,- Version 802.X for the FMS UNS-1L Software This d (...) | 24.09.2009 |
| 10028563 | DASSAULT FALCON SERVICE | P-EASA.A.S.03652 | | | MYSTERE FALCON 900 | Mode S dual transponder installation with elementary and enhanced surveillance capabilities | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Compliance has been shown in accordance with the requisites laid down by JAA Temporary G (...) | DFS Change Approval Sheet DFS 9550-CAS Issue A, dated 20th January 2010- DFS Modification Sheet DFS 9550 Issue A, dated 20th January 2010- DFS Aeroplane Flight Manual Supplement Number DFS F900-030, Original issue, dated 18th January 2010 or later re (...) | This EASA approval is applicable on those Basic Mystere Falcon 900 aircraft previously equipped with:- Dual ATC Transponders Honeywell MST-67A P/N 066-01143-0601 with antenna diversity installation- 2 IRS and 2 ADC AZ 810As per DFS Change Approval Sh (...) | 22.01.2010 |
| 10028776 | DASSAULT FALCON SERVICE | EASA.A.S.03729 | | | FALCON 900 | | | | | 20.07.2009 |
| 10029669 | DASSAULT FALCON SERVICE | P-EASA.A.S.03764 | | | FALCON 900EX | F900EX Serial Number 111 aircraft completion definition | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Compliance has been shown in accordance with the requisites laid down by EASA CRI F-002 (...) | -DFS Change Approval Sheet DFS 9565-CAS Issue A dated 30th March 2010-DFS Master Document List DFS-DE-17-FE-9665, Issue A dated 16th October 2009-DFS Modification Sheet DFS 9565, Issue A dated 29th October 2009 or later revisions of the above listed documents approved by EASA | This EASA design change airworthiness approval is applicable on Falcon 900EX serial number aircraft 111 | 12.04.2010 |
| 10032131 | DASSAULT FALCON SERVICE | | | | MF50 AND FALCON 50EX | Stretchers installation. Provide the ability to install in the aircraft one or two stretchers for air ambulance according to two configurations: One stretcher is installed in the LH side instead of the two LH seats. Two stretchers are installed in the LH side instead of the four LH seat. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | DFS Change Approval Sheet DFS5691-CAS Issue A dated 11.10.2010. DFS Modification Sheet DFS5691, Issue A dated 11.10.2010 or later revisions of the above listed documents approved by EASA | This modification is applicable on Mystere Falcon 50 and Falcon 50EX aircraft equipped with four seats installed in the LH side and two seats and a sofa installed in the RH side as defined in the Modification Sheet DFS5691. This installation does not co (...) | 12.10.2010 |
| 10032163 | DASSAULT FALCON SERVICE | | | | MF900 & F900EX | 230VAC electrical distribution for galley. An inverter and a GFI are installed to provide 230VAC power on one outlet in the galley. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- \$25X1360 & \$25X1499 at JAR 25 change 13- \$25.869 (a) (1)&(4) at JAR 25 Change 14. The (...) | DFS Change Approval Sheet DFS9621-CAS Issue A dated 12.10.2010. DFS Modification Sheet DFS9621, Issue A dated 07.10.2010 or later revisions of the above listed documents approved by EASA | Modification applicable to MF900, F900EX aircraft with :- 1400W available on A4 bus, and- Master switch in cockpit to switch off component electrically supplied in the Galley area, and- Master switch in cockpit to switch off all 115/230VAC distribution (...) | 13.10.2010 |
| 10032204 | DASSAULT FALCON SERVICE | | | | F900EX | F900EX S/N 086 Cabin Layout Modification. New interior cabin configuration features a 12-passenger and 3-crew accommodation. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- \$25.853 (a) (b)+ App F Part I and II at JAR 25 Change 14. The requirements for environ (...) | DFS Change Approval Sheet DFS9613-CAS Issue A dated 15.10.2010 DFS Modification Sheet DFS9613, Issue A dated 15.10.2010 DFS Master Data List DFS-DE-17-F9-9613, Issue A dated 11.10.2010 or later revisions of the above listed documents approved by EASA | Modification applicable to F900EX S/N 086 with a pre mod configurations described in DFS Modification Sheet DFS9613, Issue A. | 15.10.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|------------------|----------------|----------------------|-------------------------------------|--|--|---|---|---------------|
| 10033835 | DASSAULT FALCON SERVICE | P-EASA.A.S.03935 | | | FALCON 900EX | Falcon 900EX aircraft serial numbers 065 and 073 - Completion/Definition. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Dassault Falcon Service (DFS) following documents: DFS Change Approval Sheet DFS 9557-CAS Issue A dated 10th February 2011/DFS Master Document List DFS-DE-17-FE-9557, Issue A dated 21st January 21st January 2011/DFS Modification Sheet DFS 9557, Issue A (...) | This EASA design change airworthiness approval is applicable only on Falcon 900EX aircraft serial numbers 065 and 073. Associated limitation: Non commercial air transportation use. | 11.02.2011 |
| 10034125 | DASSAULT FALCON SERVICE | | | FAA STC ST02863CH | MF50, FALCON 50EX | Stretchers installation for medical evacuation. Provide the ability to install in the aircraft one or two stretchers for air ambulance operations. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | DFS Change Approval Sheet DFS5697-CAS Issue B dated 25 January 2011/DFS Modification Sheet DFS5697, Issue B dated 25 January 2011 or later revisions of the above listed documents approved by EASA. | This STC does not cover the qualification of the installed stretcher. It is to be used in conjunction with EASA STC covering the validation of FAA STC ST02863CH. This modification does not cover the use of any auxiliary medical equipment to be used in cockpit (...) | 09.03.2011 |
| 10034893 | DASSAULT FALCON SERVICE | | | | F2000, F2000EX | 115VAC electrical distribution for galley. DFSD144 A consists in installing a 28VDC/115VAC inverter, the associated protections and an outlet in the galley area. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | DFS Change Approval Sheet DFSD144-CAS Issue A dated 10.05.2011. DFS Modification Sheet DFSD144, Issue A dated 10.05.2011 or later revisions of the above listed documents approved by EASA | Modification applicable to F2000 & F2000 EX aircraft with:- 1400W available on A4 bus (version 1) or 1280W (version 2), and- Master switch in cockpit to switch off components electrically supplied in Galley area, and- Master switch in cockpit to switch (...) | 11.05.2011 |
| 10035568 | DASSAULT FALCON SERVICE | | REV. 1 | | F900EX, F900C, MF900 B | WiFi activation on Falcon 900 non Easy aircraft. This Major Change to EASA STC 10035568 approves the extension of applicability to MF900B aircraft. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- § 25.1301 (a) (d) - 1309 (a) (g) - 1322 - 1353 (a) - 1431 (a) (c) at JAR 25 Change 14. (...) | DFS Change Approval Sheet DFS9609-CAS Issue B dated 13.12.2011/DFS Modification Sheet DFS9609, Issue B dated 09.09.2011/DFS Master Data List DFS-DE-17-F9-9609, Issue B dated 02.12.2011 or later revisions of the above listed documents approved by EASA | F900EX and 900C aircraft equipped with WiFi equipments installation compliant with prerequisite note DT-003-10. MF900B aircraft equipped with WiFi equipments installation compliant with prerequisite note DT-001-11. | 22.12.2011 |
| 10037056 | DASSAULT FALCON SERVICE | | | | F900B, F900C & F900EX | Cabin Master installation. A Cabin Master Switch is installed on Co-pilot console. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | DFS Change Approval Sheet DFS9646-CAS Issue A dated 30.06.2011. DFS Modification Sheet DFS9646, Issue A dated 30.06.2011. DFS Airplane Flight Manual Supplement N° F900-031 dated 30.06.2011 or later revisions of the above listed documents approved by EASA | Not applicable | 27.10.2011 |
| 10037415 | DASSAULT FALCON SERVICE | | | | F900B, F900C, F900EX F900EX EASY | FR14 CABIN PARTITION WITH MECHANICAL POCKET SLIDING DOOR. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- § 789 (a) - 812(b)(1)(a) - 1301 - 1309 (c) - 1322 - 1353 (a)(b) - 1529 - 1541 - 1581 (...) | DFS Change Approval Sheet DFS9654-CAS Issue A dated 06.08.2011. DFS Modification Sheet DFS9654, Issue A dated 28.07.2011. DFS Airplane Flight Manual Supplement N°- F900-033 dated 27.07.2011 for F900B- F900-034 dated 27.07.2011 for F900C & F900EX- F900 (...) | As per DFS Change Approval Sheet DFS9654-CAS Issue A dated 06.08.2011. | 25.11.2011 |
| 10037417 | DASSAULT FALCON SERVICE | | | | F900EX | F900EX S/N 038 Cabin Layout Modification. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- § 25.X1499 - § 25.853 (a) (b) + Appendix F Part I (-&-&-) - II at JAR 25 Change 14 The requ (...) | DFS Change Approval Sheet DFS9655-CAS Issue A dated 10.09.2011. DFS Modification Sheet DFS9655, Issue A dated 20.10.2011 or later revisions of the above listed documents approved by EASA | Restricted to F900EX S/N 038 | 25.11.2011 |
| 10038490 | DASSAULT FALCON SERVICE | | REV. 2 | | F900EX EASY | TCAS 2000 upgrade to TCAS change 7.1 on F900EX EASY. Major Change to EASA STC 10038490. This Major Change consists in removing the former limitation (only Version 1 (F900EX EASY Step 3+) was approved) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | DFS Change Approval Sheet DFS9664-CAS Issue B dated 05.04.2012/DFS Modification Sheet DFS9664, Issue B dated 05.03.2012/DFS Master Document List No noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | As per DFS Change Approval Sheet DFS9664-CAS Issue B dated 05.04.2012 | 08.05.2012 |
| 10039335 | DASSAULT FALCON SERVICE | | | | F2000EX EASY | TCAS 2000 upgrade to TCAS change 7.1 on F2000EX EASY | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | DFS Change Approval Sheet DFSD149-CAS Issue A dated 07 March 2012/DFS Modification Sheet DFSD149 Issue A dated 07 March 2012/DFS Master Document List N° DFS-DE-17-FD-D149 Issue A dated 20 April 2012 or later revisions of the above listed documents approved by EASA | As per DFS Change Approval Sheet DFSD149-CAS Issue A dated 07 March 2012 | 26.04.2012 |
| 10041183 | DASSAULT FALCON SERVICE | | | | MYSTERE-FALCON 50 | Mode S transponder upgrade to comply with Enhanced Surveillance. Honeywell transponder MST-67A version is upgraded and ATC/TCAS Control Box is replaced by new Honeywell PS-578A unit. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | DFS Change Approval Sheet DFS5712-CAS Issue A dated July 31st (->->-) 2012. DFS Modification Sheet DFS5712, Issue A dated July 31st, 2012. DFS Airplane Flight Manual Supplement N° F50-62 dated July 31st, 2012, or later revisions of the above listed documents approved by EASA | none | 28.08.2012 |
| 10041623 | DASSAULT FALCON SERVICE | | | | F2000, F2000EX | WiFi activation on Falcon 2000, F2000EX aircraft. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. § 25.1301 (a) (d) - 1309 (a) (g) - 1322 - 1353 (a) - 1431 (a) (c) at JAR 25 Change 14. T (...) | DFS Change Approval Sheet DFS151-CAS Issue A dated 07.08.2012/DFS Modification Sheet DFS151, Issue A dated 07.08.2012/DFS Master Data List DFS-DE-17-FD-D151, Issue A dated 07.08.2012 or later revisions of the above listed documents approved by EASA | F2000 and F2000EX aircraft equipped with WiFi equipments installation compliant with prerequisite note DT-001-12. | 02.10.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|------------------|----------------|----------------------|-------------------------------------|---|--|---|---|---------------|
| 10043014 | DASSAULT FALCON SERVICE | EASA.A.S.01506 | REV. 1 | | FAN JET MYSTERE-FALCON 20 SN 397 | DFS2483A dtd. 12.Mai 2006Under Wing installation of particle measuring systems sensor | | see Change approval sheet DFS 2483A:- AFM Supp MF 20 GF 47 Revision 1 Utilisation de l'avion équipe de quatre sondes FMS sous voilure and associated list of affected technical documents | 1. This STC is limited to MF 20 GF SN 397.2. Dassault Aviation M2719 to be applied (hard points installation).3. Only symmetrical configurations are allowed (1+1 or 2+2).4. Maximum landing weight is limited to 11.616kg5. This STC is approved only for (...) | 15.05.2006 |
| 10043152 | DASSAULT FALCON SERVICE | EASA.A.S.C.01063 | | EASA.A.S.00182 | FALCON 20 EQUIPPED WITH COLLIN | Deviation to STC EASA.J.M.A.S.00237 (Rockwell Collins Proline IV DFSInstallation) lawChange approval sheet DFS2524 issueA dtd 15 July 05 | | -see Change approval sheet DFS 2524 issue AAMM chapter S DFSIDE/SF2 AX and associated list of affected technical documents | 1. This STC is applicable to Falcon 20 equipped with Collins Proline IV/Avionics only2. This STC is approved only for the configuration as defined in ModSheet DFS 2524 Issue A3. This STC is approved only for the product configuration as defined in the (...) | 17.08.2005 |
| 10043153 | DASSAULT FALCON SERVICE | EASA.A.S.C.01064 | | EASA AS00175 | FALCON 20 EQUIPPED WITH COLLIN | Installation of modified cockpit controls, lods and panels law DFSChange Approval Sheet2385 Issue A dtd 15 July 05 | | -see Change approval sheet DFS 2385 Issue AAMM Supplement DF SIDE/SB F2 - Part 76-10-00 and associated list of affected technical documents | 1. This STC is applicable to Falcon 20 equipped with Collins Proline IV/Avionics only2. This STC is approved only for the configuration as defined in ModSheet DFS 2385 Issue A3. This STC is approved only for the product configuration as defined in (...) | 17.08.2005 |
| 10043154 | DASSAULT FALCON SERVICE | EASA.A.S.C.01065 | | EASAA 800725 | MF 20 GF SIN 397 | Installation of 2 viewing panels on top of the fuselage | | -see Change approval sheet DFS 2490/AAMM no DFSIDE/S F2 AI and associated list of affected technical documents | 1. Applicable only to MF 20 SIN 3972. This STC is approved only for the configuration as defined in ModSheet DFS 2490/3 This SIC is approved only for the p/Oduct configuration as defined in the approved design data referred to in the paragraph "Descript (...) | 17.08.2005 |
| 10043156 | DASSAULT FALCON SERVICE | EASA.A.S.C.01067 | | EASA.A.S.00717 | Data not found | Installation of a 2nd Air Data System | | -see Change approval sheet DFS 2474:AFM supplement MF 20 GF N 35 to DIM 30010 and associated list of affected technical documents | 1 Applicable only to MF 20 SIN 3972. This STC is approved only for the configuration as defined in ModSheet DFS 24741 This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Descript (...) | 17.08.2005 |
| 10043318 | DASSAULT FALCON SERVICE | EASA.A.S.03728 | | | FALCON 900 | Installation of two Honeywell 12 or 24 Channels GPS sensors(GNSSU) | | - Master Document List DFS-DE-17-F9-9119 Issue H# Aeroplane Flight Manual Supplement DFS Supplement F900-029 Original version/Or later EASA approved revisions of these documents. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The ins (...) | 08.07.2009 |
| 10043502 | DASSAULT FALCON SERVICE | | | | F2000, F200EX | LIFEPORT Powered Loading System Installation Patient Lift System is installed on PAX door for boarding patients. The whole system is self-powered and independent of the A/C electrical system. It is disposable for ground use only. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | DFS Change Approval Sheet DFS153-CAS Issue A dated 31.01.2013; DFS Modification Sheet DFS153, Issue A dated 25.01.2013; or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 01.02.2013 |
| 10044032 | DASSAULT FALCON SERVICE | | REV. 2 | | MF50 | Installation of dual FMS UNIVERSAL UNS-1Lw SCN 1000.X Previous 2 main FMS UNS-1Lw with integrated GPS are replaced by two new FMS UNS-1Lw (SCN 1000.X with X >= 6) with integrated GPS / WAAS receiver. A third one with deactivated integrated GPS receive (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | DFS Change Approval Sheet DFS5687-CAS Issue B dated April 3rd, 2013. DFS Modification Sheet DFS5687, Issue B dated April 3rd, 2013. DFS Airplane Flight Manual Supplement N°F50-66 dated April 3rd, 2013. or later revisions of the above listed documents approved by EASA | As per DFS Change Approval Sheet DFS5687-CAS Issue B dated April 3rd, 2013 and DFS Airplane Flight Manual Supplement N°F50-66 dated April 3rd, 2013. This STC revision (or later revisions approved by EASA) supersedes EASA approval STC 10044032 rev. 0, da (...) | 04.04.2013 |
| 10044033 | DASSAULT FALCON SERVICE | | | | MF50 | P-RNAV Capability To demonstrate compliance with TGL N°10 Rev 1 concerning P-RNAV operations "Airworthiness and operational approval for Precision RNAV operations in designated European airspace" | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -DFS Change Approval Sheet DFS5659-CAS Issue B dated March 08th, 2013. -DFS Modification Sheet DFS5659, Issue B dated March 08th, 2013. -DFS Airplane Flight Manual Supplement N°F50-61 dated March 08th, 2013 (for applicability on A/C with DFS 5687 dedic (...) | As per DFS Change Approval Sheet DFS5659-CAS Issue B dated March 08th, 2013. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or (...) | 13.03.2013 |
| 10044034 | DASSAULT FALCON SERVICE | | | | MF50 | COMPLIANCE WITH AC 90-100A to demonstrate compliance with the RNAV 1 and RNAV 2 regulations described in ICAO document 9613 AN/937 "Performance-based Navigation (PBN) Manual" for Falcon 50 aircraft equipped with FMS UNS-1Lw software 1000.X (X#6) according to DFS 5687 issue A or after. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | DFS Change Approval Sheet DFS5688-CAS Issue A dated March 08th, 2013. DFS Modification Sheet DFS5688, Issue A dated March 08th, 2013. DFS Airplane Flight Manual Supplement N°F50-63 dated March 08th, 2013. or later revisions of the above listed documents approved by EASA | As per DFS Change Approval Sheet DFS5688-CAS Issue A dated March 08th, 2013. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or (...) | 13.03.2013 |
| 10044035 | DASSAULT FALCON SERVICE | | | | MF50 | RNP-10 and RNP-4 capability to demonstrate compliance with the RNP-10 and RNP-4 regulations for Falcon 50 aircraft equipped with FMS UNS-1Lw software 1000.X (X#6) according to DFS 5687 issue A or after. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | DFS Change Approval Sheet DFS5689-CAS Issue A dated March 08th, 2013. DFS Modification Sheet DFS5689, Issue A dated March 08th, 2013. DFS Airplane Flight Manual Supplement N°F50-64 dated March 08th, 2013. or later revisions of the above listed documents approved by EASA. | As per DFS Change Approval Sheet DFS5689-CAS Issue A dated March 08th, 2013. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or (...) | 13.03.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|-------------------|----------------|--------------------------------|---|---|--|--|--|---------------|
| 10044036 | DASSAULT FALCON SERVICE | | | | MF50 | RNP APP Capability/Description as per Associated Technical Documentation | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | DFS Change Approval Sheet DFS5690-CAS Issue A dated March 08th, 2013. DFS Modification Sheet DFS5690, Issue A dated March 08th, 2013. DFS Airplane Flight Manual Supplement N°F50-65 dated March 08th, 2013. Prior to installation of this design change it must be determined that the interrelationship between th (...) | As per DFS Change Approval Sheet DFS5690-CAS Issue A dated March 08th, 2013 and DFS Airplane Flight Manual Supplement N°F50-65 dated March 08th, 2013. Prior to installation of this design change it must be determined that the interrelationship between th (...) | 04.04.2013 |
| 10015422 | DAVIS AERO ENGINEERING INC. | EASA.IM.A.S.01153 | | FAA STC ST01657LA | LEARJET 60 | Installation of Rockwell Collins TCAS II. | | FAA STC ST01657LA. Davis Aero Engineering MDL Doc. No AG362-MDL, Rev. B, or later FAA-approved revisions. AFMS AG362-600, Rev. IR, or later FAA-approved revisions. | 1. For certification basis refer to FAA TCDS A10CE plus JAATGLs No 8 and 13. | 01.04.2005 |
| 10015747 | DAYTON-GRANGER Inc. | EASA.IM.A.S.01779 | REV. 4 | FAA STC NO. ST02241NY | CL-600-2B19 CL-600-2C10 CL-600-2D15 CL-600-2D24 CL-600-2E25 | Installation of Winglet and Horizontal Stabilizer End Cap Lightning Protection Kits | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged (...) | EASA STC No. 10015747 Revision 3 dated 19/02/2010. FAA STC No. ST02241NY dated 18 March 2011. Configuration 1 (CL-600-2B19): - Dayton Granger Master Drawing List P/N DCS740070-(-) dated February 7, 2007 or later approved revisions. - Dayton Granger Instru (...) | None | 24.06.2011 |
| 10016043 | DC AEROSPACE | EASA.IM.A.S.02214 | REV. 1 | ST03011AT | BOEING 727-100 SERIES BOEING 727-200 SERIES | Installation of Sandel ST3400 Terrain Awareness and Warning System (TAWS) in Boeing 727 Series Aircraft according to FAA STC ST03011AT dated 22 December 2004 | | - MDL-2418, Revision H dated 25 July 2005- AFM Supplement FMS-2418-1, Rev. IR dated 23 December 2004- Instructions for Continued Airworthiness ICA-2418, Rev. B dated 17 December 2004 or later revisions of the above listed documents approved by EASA in (...) | SANDEL mandatory SB 3400-09 to be incorporated. The limitations as listed in FAA STC ST03011AT apply. | 23.07.2007 |
| 10015830 | DECA AVIATION ENGINEERING LIMITED | EASA.IM.A.S.01919 | REV 2 | TCCA LSTC O-LSA06-056/D ISS1 | DHC-6-300 | Installation of an ASIRAS Radar System in a DHC-6-300 aircraft in accordance with Modification Summary for ASIRAS Radar System Installation. DECA Aviation Engineering Limited document DCP 05259, Revision 2, dated 14/12/05 or later EASA approved revision | | Data package containing the following documents, and additional supporting data: 1. Design Compliance Program for ASIRAS Radar System Installation. DECA Aviation Engineering Limited document DCP 05259, Revision 12. Static Strength and Compliance Requi (...) | 1. DECA Aviation Flight Manual Supplement FMS05259, Revision 1, is required | 18.07.2007 |
| 10016112 | DECA AVIATION ENGINEERING LIMITED | EASA.IM.A.S.02302 | | TCCA STC O-LSA02-397/D ISS 2 | A320-214 | Installation Of 180 Passenger - All Economy Class Seating Arrangement, in accordance with Transport Canada STC O LSA02-397/D Issue 2 | | Master Drawing List, Document Number MS02071 Revision 1, dated 18th July 2002 or later EASA approved revision. Instructions for Continued Airworthiness, Document Number ICA02071, Revision 1, dated 18th July 2002, or later EASA approved revision. | 1. As per Transport Canada STC O-LSA02-397/D Issue 2.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compati (...) | 10.05.2007 |
| 10016342 | DECA AVIATION ENGINEERING LIMITED | EASA.IM.A.S.02656 | REV. 1 | TCCA SA00-98 ISSUE 4 | BD-700-1A10 | Baggage Door Locking System Modification. | | Transport Canada STC SA00-98 Issue 4, dated 3 October 2007. Transport Canada Statement of Compliance reference 5010-O-07-0698, dated 8 May 2008. DECA Aviation Modification Summary No. MS0009938, Section 1, Issue B, dated 24 November 2000 or later approved revisions. | | 23.04.2008 |
| 10016431 | DECA AVIATION ENGINEERING LIMITED | EASA.IM.A.S.02771 | REV 1 | TCCA STC O-LSA07-260/D ISSUE 2 | BOMBARDIER BD-700-1A10 | Cabin Interior Reconfiguration. | | TCCA STC No. O-LSA07-260/D Issue 3 dated 18 August 2008. TCCA SoC reference 5010-O-08-0543 dated 06 November 2008. DECA Aviation Modification Summary MS07158 revision 2 or later approved revision. DECA Instructions for Continued Airworthiness ICA07158 R (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 12.11.2008 |
| 10016482 | DECA AVIATION ENGINEERING LIMITED | EASA.IM.A.S.02836 | | TC STC O-LSA08-069/D | BOMBARDIER DHC-6-300 | Radar System Installation. | | Installation in accordance with DECA Aviation Engineering Ltd Modification Summary MS 07099 Rev 1 or later Transport Canada approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 12.11.2008 |
| 10027141 | DECA AVIATION ENGINEERING LIMITED | | REV. 1 | TCCA SA00-126; CAA NL SA0014NL | DHC-8-102, -103, -106 DHC-8-201, -202 DHC-8-301, -311, -314, -315 | QAR/ASAP Monitoring System TCCA SA00-126, Is. 4 | FAR Part 25 as per TCDS EASA.IM.A.191 for DHC-8-100/200/300 Series | A) Models: DHC -8- 102, -103, -106 Installation must be performed in accordance with DECA Aviation Modification Summary MS000952, Revision 12 *)<(-> This modification must be maintained in accordance with DECA Maintenance Manual Supplement DECA-4903- (...) | The QAR system configurations are compatible with each of the following FDR/FDAU combinations: For Model DHC -8-102, -103, -106, -201, -202L3 FDR No: 2100-4043-00 with Plessey FDAUNo: 612/1/49780/102 (Model PA3810B1/Post Mod 8/1876). For Model DHC -8 - (...) | 04.01.2010 |
| 10029659 | DECA AVIATION ENGINEERING LIMITED | | | TCCA LSTC O-LSA09-37/D | B737-800 | Interior Reconfiguration & miscellaneous Cabin Alterations /Modification and relocation of the passenger seats along with relocation/installation of emergency equipments. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | TCCA LSTC Number: O-LSA09-37/D Issue 1 dated Feb 03, 2009- DECA Modification summary MS08165 dated Feb 03, 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision) | Valid for MSN 30292 only | 09.04.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|--------------------|----------------|-----------------------------|--|--|--|---|--|---------------|
| 10029922 | DECA AVIATION ENGINEERING LIMITED | | | TCCA LSTC O-LSA09-040/D | B737-800 | Sliding Carpet Loading System Removal/Deactivation | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TCCA LSTC O-LSA09-040/D, dated Feb 05, 2009-DECA Modification Summary MS09024 Rev.3 dated Apr 29, 2009-ICA 09024 Rev.2 dated Mar 11, 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision) | Valid for MSN30292 and 33029 only | 03.05.2010 |
| 10030614 | DECA AVIATION ENGINEERING LIMITED | | | TCCA STC O-LSA06-007/D | B737-800 | Passenger Seating Arrangement - Installation / Relocation of passenger Seats in 186-189 Pax Configuration and installation of Emergency Equipments | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. | DECA Modification Summary No. MS05289 Rev.16 dated 06 January 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision) | Only valid for MSN 28373, 28375, 29351, 29642, 29883, 30637, 30671, 33003, 33029, 33699, 34704, 35093 | 25.06.2010 |
| 10032064 | DECA AVIATION ENGINEERING LIMITED | | | TCCA STC O-LSA10-43 | DHC-6-300 | Radar Antenna Installation - Structural Provisions i.a.w. TCCA STCO-LSA10-043/D | The Certification Basis for the original product remains applicable to this certificate/ approval. | -See TCCA STC O-LSA10-043/D, issue 1-DECA Aviation Engineering Ltd. Modification Summary MS09082 Rev. 1-DECA Aviation Engineering Ltd. Flight Manual Supplement SDC-AFMS-0703 Rev. NCPlacard (see Drwg. No. 9003717) in view of pilot: "Aircraft not to beo (...) | a) For serial number 235 only. b) Flight in known icing conditions are not approved. | 06.10.2010 |
| 10033224 | DECA AVIATION ENGINEERING LIMITED | | | O-LSA07-225/D, ISSUE 1 | P-180 AVANTI II | TCA STC No. O-LSA07-225/D Issue 1 "Import Configuration - Interior Completion (including various cabin installations)" | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AFM Supplement No. FMS10131.1, Rev. 1 dated 7 December 2010 Weight & Balance Manual Supplement No. WBS10131, Rev. 1 dated 7 December 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision) | None | 04.01.2011 |
| 10038463 | DECA AVIATION ENGINEERING LIMITED | | | TCCA STC NO. O-LSA09-344/D | CL-600-2B19 | Indium Antenna Installation. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. O-LSA09-344/D Issue 1 dated 19 November 2009, DECA Aviation Engineering Drawing No. 5002572, Revision A or later approved revisions; DECA Aviation Engineering Instructions for Continued Airworthiness ICA09215, Revision 1 or later (...) | Restricted to ASN 8092. | 17.02.2012 |
| 10040912 | DECA AVIATION ENGINEERING LIMITED | | REV. 1 | TCCA STC NO. O-LSA00-829/D | BD-700-1A10 | Baggage Door Locking System Modification | EASA, IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged (...) | Transport Canada STC No. O-LSA00-829/D Issue 2 dated 04 January 2002, DECA Modification Summary MS000938 Issue 1 dated 08 November 2000, DECA Maintenance Manual Supplement DECA-4885-02 Revision 2, TC accepted 19 December 2001 or later revisions of the (...) | Restricted to ASN 9003 | 28.08.2012 |
| 10043385 | DECA AVIATION ENGINEERING LIMITED | | REV. 1 | TCCA SH11-47 ISSUE 2 | AB/AW 139 | TCCA STC SH11-47 Issue 2 "NICAD to Concorde Lead-Acid Battery Replacement". | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | DECA Aviation Engineering Flight Manual Supplement FMS11143-01 Revision 2 (for helicopters equipped with RG-380E/44L main battery); DECA Aviation Engineering Flight Manual Supplement FMS11143-02 Revision 2 (for helicopters equipped with RG-380E/60L mai (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 14.02.2013 |
| 10015489 | DECA AVIATION ENGINEERING LTD | EASA, IM.A.S.01251 | | - | A321-211, SN 675 | Interior reconfiguration 20J/152 Y seating arrangement | | | 1. This STC is limited to A321 MSN 675.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.07.2005 |
| 10015492 | DECA AVIATION ENGINEERING LTD | EASA, IM.A.S.01254 | | TCCA STC O-LSA04-008/D | A320-232 | Passenger Seat Cushion Assembly Replacement on B/E Aerospace, seat typ: D3111F11xx-01 series | | Section 1.0 of DECA Modification Summary No. MS03186, revision 1, dated January 27, 2004 or later EASA-approved revision | 1. This STC is limited to the A320 MSN 496. | 18.10.2005 |
| 10015493 | DECA AVIATION ENGINEERING LTD | EASA, IM.A.S.01255 | | TCCA STC O-LSA05-018/D | A320-232 | Reconfiguration of existing 180 passenger all economy seating arrangement | | DECA Aviation Engineering Modification Summary MS05017, Revision 1, dated January 20, 2005 or later EASA approved revision | 1. This STC is limited to the A320 MSN 496.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.10.2005 |
| 10015494 | DECA AVIATION ENGINEERING LTD | EASA, IM.A.S.01256 | | TCCA STC O-LSA02-327/D | A319-112, A320-211, A320-212, A320-214, A320-231, A320-232 | Flight Warning Computer Altitude Callout Reconfiguration | | Avionics Design Services Ltd. Engineering Order EO02001-1, Revision 7, dated October 14, 2004 or later EASA approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.10.2005 |
| 10015587 | DECA AVIATION ENGINEERING LTD | EASA, IM.A.S.01498 | | TCCA STC O-LSA01-428/D ISS3 | A320-232 | Emergency exit sign Relocation | | DECA Drawing 7SK0439, Revision A, dated April 5, 2001 or later EASA approved revision | 1) Limited to the A320 MSN 4962. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.10.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|---------------------|----------------|-------------------------------|---|--|---|---|--|---------------|
| 10015791 | DECA AVIATION ENGINEERING LTD | EASA.IM.A.S.01849 | | TCCA LSTC O-LSA02-653/D ISS 3 | A318-112, A320-214 | Passenger seat leather dress cover replacement | | DECA Modification summary n° MS02136, revision 3, dated December 22, 2003 or later EASA approved revision | This STC is limited to: A319 MSN 1846, A319 MSN 1598 and A320 MSN 1686 | 12.05.2006 |
| 10042564 | DECA AVIATION ENGINEERING LTD | EASA.IM.A.S.01043 | REV. 1 | | A320-200 | 174 Passenger - All Economy Class Passenger Seating Arrangement,O-LSA04-318/D iss 03 | | Instructions for Continued Airworthiness 04230-02 Rev 1-Instructions for Continued Airworthiness 04230-03 Rev 1-Instructions for Continued Airworthiness 04230-04 Rev 1 | 1. This STC is limited to aircraft S/N 645, 579 and 525.2. This STC is approved only for the product configuration as defined in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 05.02.2013 |
| 10016241 | DECRANE AEROSPACE PATS AIRCRAFT, | EASA.IM.A.S.02513 | REV. 1 | FAA STC ST00741LA-D | BOEING 737-700 | Installation of Canadian Marconi CMA-2101 High Gain SATCOM Antenna.This STC revises and supersedes EASA STC EASA.IM.A.S.02513. The revision was issued to reflect the transfer from Hollingshead International to Decrane Aerospace PATS Aircraft. | The Certification Basis for the original product remains applicable to this certificate/ approval. | Hollingshead Master Data List 12094 Rev. C dated July 02, 1999 or later revision of the above document approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision). | None. | 18.11.2009 |
| 10016649 | DECRANE AEROSPACE PATS AIRCRAFT, | EASA.IM.A.S.03093 | REV. 1 | FAA STC ST02057LA | BOEING 737-700 SERIES BOEING 737-800 SERIES | Installation of Electronic Equipment Rack in Aft Bulk Cargo Compartment.This STC revises and supersedes EASA STC EASA.IM.A.S.03093. The revision was issued to reflect the transfer from Hollingshead International to Decrane Aerospace PATS Aircraft. | The Certification Basis for the original product remains applicable to this certificate/ approval. | Hollingshead Master Data List 2878-01 Rev. A dated June 07, 2007/Instructions for Continued Airworthiness 2878-14 Rev. A dated June 14, 2007 or later revision of the above document approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision). | None. | 18.11.2009 |
| 10026943 | DECRANE AEROSPACE PATS AIRCRAFT, | P-EASA.IM.A.S.03257 | | | B737-800 | Installation of Auxiliary Fuel System in the forward and aft cargo compartments.Validation of STC ST01713NY-D as amended June 29, 2009 | In accordance with TCDS | Configuration 1: Installation of a Four-Cell Auxiliary Fuel System in the forward and aft cargo compartment PATS Aircraft LLC Kit Installation Drawing 5140-2811-103 as listed in the Master Drawing List 5140-2900-001, revision B dated February 6, 2006. Airplane Flight Manual Supplement No. 132 for a Four- | None. | 26.08.2009 |
| 10027297 | DECRANE AEROSPACE PATS AIRCRAFT, | EASA.IM.A.S.03026 | REV. 4 | FAA STC ST01727NY-D | ERJ 190-100 ECJ | Installation of an Executive Interior Major Change to FAA STC ST01727NY-D Update of EASA STC 10027297 Rev 3: Compliance with EWIS ICA Special Condition H-01 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI A-1 IM.A.S.03026 The requirements for environmental protection and the associated certified noise and/or emissions lev (...) | Configuration 1:- Master Drawing List 5950-0000-MDL Rev. G dtd. 16.April 2009 (*) - Airplane Flight Manual Supplement No. 215, Rev. D dtd. 12. March 2010 (*) - Maintenance Manual Supplement TO-3163 Rev. IR dtd. 1.May 2008 (*) with TR20-1001 dtd. 10.April 2009 (*) | Configuration 1:- Concurrent installation of STC ST01728NY-D (Class "C" Cargo Compartment) iaw. Master Drawing List 6071-0000-MDL Rev. IR dtd. 7.October 2008 is required. - Commercial operation under EASA rules requires installation of PATS SBST01728NY-D (...) | 24.06.2013 |
| 10027298 | DECRANE AEROSPACE PATS AIRCRAFT, | EASA.IM.A.S.03027 | REV. 2 | FAA STC ST01728NY-D | ERJ 190-100 ECJ | Modification of the Class C Forward Cargo Compartment to accommodate an Auxiliary Fuel Tank and Installation of a new Class C Cargo Compartment at the Rear of the Main Cabin Major Change to FAA STC ST01728NY-D Update of EASA STC 10027298 Rev 1: Compliance with EWIS ICA Special Condition H-01 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI A-1 IM.A.S.03026 The requirements for environmental protection and the associated certified noise and/or emissions lev (...) | Configuration 1: Modification of the Class C Forward Cargo Compartment to accommodate an Auxiliary Fuel Tank and Installation of a new Class C Cargo Compartment at the Rear of the Main Cabin. - Master Drawing List 6071-0000MDL Rev. IR dtd. 7.October 2008 (...) | Configuration 1:- Concurrent installation of STC ST01727NY-D (Installation of an Executive Interior) iaw. Master Drawing List 5950-0000-MDL revision G dtd. 16.April 2009 is required. ----- Configuration 2:- Concurrent i (...) | 24.06.2013 |
| 10027299 | DECRANE AEROSPACE PATS AIRCRAFT, | EASA.IM.A.S.03028 | REV. 2 | FAA STC ST01729NY-D | ERJ 190-100 ECJ | Installation of a Passenger Gaseous Oxygen Supply System Major Change to FAA STC ST01729NY-D Update of EASA STC 10027299 Rev 1: Compliance with EWIS ICA Special Condition H-01 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI A-1 IM.A.S.03026 The requirements for environmental protection and the associated certified noise and/or emissions lev (...) | Master Drawing List 6072-0000MDL Rev. A dtd. 15.December 2008 (*) - Airplane Flight Manual Supplement No. 226, Rev. IR dtd. 17.April 2009 (*) - Maintenance Manual Supplement TO-3163 Rev. IR dtd. 1.May 2008 (*) with TR20-1001 dtd. 10.April 2009 (*) or la (...) | Concurrent installation of STC's ST01727NY-D (Installation of an Executive Interior) and ST01728NY-D (Class "C" Cargo Compartment) is required. For aircraft under FAA jurisdiction this modification may not be installed after 30.August 2010 unless: (1) (...) | 24.06.2013 |
| 10028092 | DECRANE AEROSPACE PATS AIRCRAFT, | | | FAA STC ST00405LA-D | 737-700 IGW | Alteration of Fire Protection Systems, Lower Cargo Compartments.This STC revises and supersedes LBA STC LBA.21NE1.TA0682. The revision was issued to reflect the transfer from Hollingshead International to Decrane Aerospace PATS Aircraft. | The Certification Basis for the original product remains applicable to this certificate/ approval.CRI A-01, B737-700, Issue 9, 13 February 1998. | Hollingshead Master Data List 12106 Rev. M dated August 01, 2002/FAA Airplane Flight Manual Supplement 20629 Rev. N/Instructions for Continued Airworthiness 20689 Rev. Cor later revision of the above document approved by EASA in accordance with EASA ED (...) | None. | 18.11.2009 |
| 10028093 | DECRANE AEROSPACE PATS AIRCRAFT, | | | FAA STC ST01184LA | B737-800 | Alteration of Fire Protection Systems, Lower Cargo Compartments.This STC revises and supersedes LBA STC LBA.21NE1.TA0511. The revision was issued to reflect the transfer from Hollingshead International to Decrane Aerospace PATS Aircraft. | The Certification Basis for the original product remains applicable to this certificate/ approval. | Hollingshead Master Data List 2554-01 Rev. E or later revision of the above document approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision). | None. | 18.11.2009 |
| 10015980 | DEL-AIR | EASA.IM.A.S.02132 | | FAA STC SA5733SW | CESSNA 150 CESSNA 150A CESSNA 150B CESSNA 150C CESSNA 150D CESSNA 150E CESSNA 150F CESSNA 150G CESSNA 150H CESSNA 150L CESSNA 150M CESSNA 152 CESSNA A152 | Installation of Long Range Wing Fuel Tanks | | Custom Aircraft Conversions Inc. Drawing List No. C1502-000, revised 05/10/87 and Revision B, dated 02/24/96, or later approved revisions. | 1. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.2. This STC is approved only for the product configuration as defined in t (...) | 14.08.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|---------------------------|--|--|---------------------|--|--|---------------|
| 10015984 | DEL-AIR | EASA.IM.A.S.02136 | | FAA STC SA01388SE | CESSNA 170A CESSNA 170B CESSNA 172 CESSNA 172A CESSNA 172B CESSNA 172C CESSNA 172D CESSNA 172E CESSNA 172F CESSNA 172G CESSNA 172H CESSNA 172I CESSNA 172J CESSNA 172K CESSNA 172L CESSNA 172M CESSNA 172N CESSNA 172P | Installation of a 7 Gallon extended range tank in each wing. | | Manufactured and installed in accordance with Data List No 11000-6, Rev IR, dated December 23, 2003, Data List No 11000-1, Rev IR, dated December 23, 2003 and Installation Instructions DA Report No 152, Rev N/C, dated September 12, 2002, or later ap (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.09.2006 |
| 10015503 | DELASTEK AERONAUTIQUE INC. | EASA.IM.A.S.01266 | | TCCA STC SA04-116 ISSUE 1 | BOMBARDIER CL-600-2B19 (SERIES) | Installation of Delastek Bleed Lead Monitoring System p/n DK-329-001. | | Transport Canada STC SA04-116 Issue 1 dated November 2004 or later approved revision. Delastek Inc. Master Document List/MDL-DK-329-001 Revision 1 dated 25 October 2004 or later approved revision. | For s/n 7003 through 7212 Bombardier Service Bulletin 601R-30-022 Part C must be integrated prior to the installation. | 23.07.2007 |
| 10015800 | DELTA AIR LINES, INC | EASA.IM.A.S.01864 | | SA3305SO | BOEING 767-200 | Installation of a Delta modified forward trunnion bearing housing & a Delta manufactured bearing retainer bolt to allow forced lubrication of the respective joints. | | Delta E.O. 1-65641-3 dated August 08 1992 or later approved revision. | | 01.06.2006 |
| 10015363 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.00677 | REV. 1 | FAA STC ST01523SE | BOEING 747-400 | 1. Configuration 1: Installation of Interior Components in accordance with Delta Engineering Master Data List 0107-05983-2128, Revision A, dated March 26, 2007 or later DAS approved. 2. Configuration 2: Installation of Interior Components in accordance with Delta Engineering Master Data List 0107-05983-2128, Revision A, dated March 26, 2007 or later DAS approved. 2. Configuration 2: Installation of Interior Components in accordance with Delta Engineering Master Data List 0107-05983-2128, Revision A, dated March 26, 2007 or later DAS approved. | | 1. 707-05808-2128 Rev - May 1, 2006 ICA for Cabin Interior Update. 2. 707-05888-2128 Rev - Sept. 8, 2006 ICA for Cabin Interior Update. 3. AMM supplement HTM-EG744-MM01 Rev. O Interior installation H9300-1, -3. | 1. The approval holder shall fulfill the obligations of Part 21, Paragraph 21.AJ.09.2, Configuration 1 - This STC must be installed concurrent with Jamco America ST01455SE, EJM Aerospace Services ST03078AT, STG Aerospace ST02993AT and Boein (...) | 09.09.2008 |
| 10015393 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.01094 | REV. 2 | FAA STC ST01507SE | BOEING 747-400 | Description of Design Change: Installation of Interior Equipment: 1. Configuration Installation of interior components in accordance with Heath Techna, Inc. Drawing List HPD-DL-55, Rev. BF, dated February 11, 2005 or later approved revision. 2. Configuration (...) | | Heath Techna Inc. Drawing List HPD-DL-55, Rev BF dated February 11, 2005 or later approved revision. 1.0107-05791-2127 Delta Engineering Master Data List 0107-05879-2127 Master Data List P94 Power Panel 3. Instruction for Continued Airworthiness docu (...) | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. 2. Configuration (...) | 01.11.2007 |
| 10015421 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.01151 | | FAA STC ST01895NY | DC-9-87 (MD-87) MD-88 | Validation of FAA STC No ST01895NY. Replacement of Dual Transponders with ACSS 7517800-10005 ATDL Transponders with ELS and Gobles G7490-58 Control Head. | | -FAA STC No ST01895NY - Delta Engineering MDL 0158-05307-2076 Rev. A dated 30 April 2004. | 1. This installation does not include EHS or ADS-B functionality. The Delta Engineering MMS No 0758-05312-2076 dated 19 April 2004 is required. | 22.07.2005 |
| 10015443 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.01183 | | FAA STC ST01310NY | AEROSPATIALE ATR 42-200 AEROSPATIALE ATR 42-300 AEROSPATIALE ATR 42-320 AEROSPATIALE ATR 42-500 AEROSPATIALE ATR 72-101 AEROSPATIALE ATR 72-102 AEROSPATIALE ATR 72-201 AEROSPATIALE ATR 72-202 AEROSPATIALE ATR 72-211 AEROSPATIALE ATR 72-212 AEROSPATIALE ATR 72-212A | Configuration 1: Installation of Delta Engineering sensors, brackets and wiring harnesses in accordance with Delta Engineering Master Drawing List 0101-04808-2030 revision F dated August 5, 2004 Configuration 2: Installation of the SFIM Inc. Auxiliary (...) | | Delta Engineering Instructions for Continued Airworthiness document number 0701-04812-2030, revision D dated June 24, 2004 for configuration 1 and 2 | 1. Configuration 1: this approval is for structural provisions only. 2. Configuration 1 is required for Configuration 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Descri (...) | 30.05.2005 |
| 10016039 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.02208 | REV. 1 | FAA STC ST02223NY-D | BOEING 767-200 | Heath Techna NuLook Interior | | FAA STC ST02223NY-DMDL 0144-05988-2136 Rev c 12/07/2007. | | 17.09.2007 |
| 10016094 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.02283 | | FAA STC ST02227NY-D | AIRBUS A340-300 | Panasonic Aviation Corporation System 3000 IFE Installation /Installation of a Panasonic Avionics Corporation System 3000 In-Flight Passenger Entertainment System with Astronics Empower @ In-Seat Power Supply System in accordance with Delta Engineering (...) | | FAA STC ST02227NY-D Master data List, Ref. No. 0152-05938-2135, Panasonic Document MDL10497, Rev. A, dated June 18th 2007. Certification Plan #New STC Panasonic System 3000 in a Airbus A340-300 Series, Ref. DAS-2135-01, Rev. E, dated March 2007 (...) | Delta Engineering Instructions for Continued Airworthiness, Ref. No. 0752-05945-2135 Revision A, dated March 8th 2007. Delta Engineering Airplane Flight Manual Supplement, Ref. No. 0652-06023-2135, dated June 21st 2007. Cabin configurat (...) | 05.07.2007 |
| 10016134 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.02330 | | FAA ST01767NY | BOEING 747-200 | Update of seat components for System 2000E per DAS project DAS-2056-03 on a B747-400. | | Data package containing the following documents, and additional supporting data: 1 DAS-2056-03 Rev. B Project plan for an update of Panasonic System 2000E in a B747-2 IS 60461 Seat equipment configuration 3 WD60461 Seat equipment wiring diagram 4 TL105 (...) | This STC is approved only for the product configuration as defined in the approved design data. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.03.2008 |
| 10016280 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.02572 | | FAA STC ST02393NY-D | BOEING 747-400 | Installation of Heath Techna Cabin Interior | | Master data list 0107-06068-2152, Rev A, and instructions for Continued Airworthiness 0707-06050-2152, Rev A, or later approved revisions. | 1. TGL 12 complied with. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the a (...) | 09.10.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|---------------------|----------------|----------------------|-----------------------|---|---|--|--|---------------|
| 10016318 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.02624 | REV. 1 | FAA STC ST02404NY-D | B767-300 | Installation of Panasonic Avionics Overhead In-Flight Entertainment System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate as defined in the TCDS plus the following for changed areas: FAR Section, Title, At Amendment, §25.8 (...) | MDL 0144-06099-2153, revision F, 08/10/2010 ICA 0744-06105-2153, Rev Cor later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | Limited to installation on aircraft already fitted with Heath Technica NLock interior (Configuration 4) ref EASA STC IM.A.S.02474 | 11.02.2011 |
| 10016329 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.02636 | | FAA STC ST02630NY | BOEING 777-200 | Heath Technica Interior | | FAA STC ST02630NY Delta Engineering Master Data List 0143-06294-2159 Revision A dated November 25, 2008 or later approved revisions | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 15.12.2008 |
| 10016496 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.02858 | | FAA STC ST02403NY-D | BOEING 747-400 | Installation of Seat In-Flight Entertainment Components and Software update in accordance with Delta Engineering Master Data List 0107-06195-2172 Revision A, dated 4 April 2008. Validation of FAA STC ST02403NY-D. | | 0107-06195-2172 Rev. A Master Data List for Panasonic Avionics IF System 3000.44-25-72 Aircraft Maintenance Manual Supplement B747-4000707-06201-2172 Rev. A Instruction for Continued Airworthiness for System 3000 IFE installation in B747-400. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 14.08.2008 |
| 10016540 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.02917 | | FAA STC ST02402NY-D | AIRBUS A320 | Panasonic IFE aFX Installation of Panasonic eFX system EAS Validation of FAA STC No. ST02402NY-D from 28. March 2008. | | FAA STC No. ST02402NY-D, from 28. March 2008 Delta Engineering Master Data List 0164-06179-2165 Revision -, from 15. March 2008, applicable for A320-200 series or later revisions of the above listed documents approved by EASA in accordance with EASA ED (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.1 (...) | 16.07.2008 |
| 10016675 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.03182 | | FAA STC ST02215NY-D | BOEING 747-400 SERIES | Installation of Panasonic Avionics Corporation System 3000 In-Flight Entertainment System in accordance with Delta Engineering Master Data List 0107-05809-2121 Rev. A dated Nov. 15, 2006 or later approved revisions Validation of FAA STC ST02215NY-D | | Delta Engineering 0707-05814-2121 Rev. A Instructions for Continued Airworthiness | 1. Delta Engineering Instructions for Continued Airworthiness 0707-05814-2121 Rev. A dated Aug 31, 2006 are required with this installation.2. The cabin configuration of the seats installed in accordance with KLM LOPA KLM3H112780 Revision ORG is required (...) | 18.06.2009 |
| 10027744 | DELTA ENGINEERING CORPORATION | P-EASA.IM.A.S.03198 | REV. 1 | FAA STC ST02643NY-D | A340-300 | Update of a Panasonic System 2000E In-Flight Passenger Entertainment System to a System 2000E Hybrid Major Change to the EASA STC 10027744 to remove EWIS limitation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01/Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane Systems (...) | -FAA STC ST02643NY-D dated 19th September 2009-Delta Engineering Master Data List, ref. doc. No. 0152-06526-2199, Revision B, dated 19th September 2009- Compliance Summary Report, ref. doc. No. 6652-06713-2199, Revision A, dated 21st October 2009-Subst (...) | None | 04.04.2012 |
| 10027745 | DELTA ENGINEERING CORPORATION | P-EASA.IM.A.S.03199 | REV. 1 | FAA STC ST02642NY-D | A330-200 | Update of a Panasonic System 2000E In-Flight Passenger Entertainment System to a System 2000E Hybrid Major Change to the EASA STC 10027746 to remove EWIS limitation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01/Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane Systems (...) | -FAA STC ST02642NY-D dated 3rd August 2009-Delta Engineering Master Data List, ref. doc. No. 0154-06517-2198, Revision A, dated 1st August 2009- Compliance Summary Report, ref. doc. No. 6600-06711-2198, Revision A, dated 21st October 2009- Substantiation (...) | The STC holder shall provide by 7th June 2010 Instructions for Continued Airworthiness on Electrical Wiring Interconnection System - EWIS ICA - in accordance with Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS". | 04.04.2012 |
| 10027746 | DELTA ENGINEERING CORPORATION | P-EASA.IM.A.S.03200 | REV. 1 | FAA STC ST02642NY-D | A330-200 | Update of a Panasonic System 2000E In-Flight Passenger Entertainment System to a System 2000E Hybrid Major Change to the EASA STC 10027746 to remove EWIS limitation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01/Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane Systems (...) | -FAA STC ST02642NY-D dated 13th July 2009-Delta Engineering Master Data List, ref. doc. No. 0154-06517-2198 (Configuration 2), Revision -, dated 10th July 2009- Compliance Summary Report, ref. doc. No. 6600-06711-2198, Revision A, dated 21st October 2 (...) | None | 04.04.2012 |
| 10029011 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.03196 | | FAA STC ST01483NY | BOEING 747-400 SERIES | | | | | 23.07.2009 |
| 10029012 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.03197 | | FAA STC ST01519NY | BOEING 747-400 SERIES | | | | | 23.07.2009 |
| 10029013 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.03195 | | FAA STC ST01482NY | BOEING 747-400 SERIES | | | | | 21.07.2009 |
| 10030011 | DELTA ENGINEERING CORPORATION | | REV. 1 | FAA STC ST02654NY-D | B757-300 | Installation of Panasonic Avionics Corporation System Digital MPES II Passenger Entertainment System. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate as per the TCDS and Special Condition H-01 "EWIS-The requirements for environmental protection and (...) | Delta Engineering Master Data List 0145-06759-2212, Revision D, 09/12/2010 Aeroplane Flight Manual Supplement, 0645-06766-2212, 15/03/2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | Finnair Minor Modifications (EASA.21J.263) EO 15330FR 1 and EO 15383 FR must be installed prior to or concurrent with the installation of this STC. | 25.01.2011 |
| 10031836 | DELTA ENGINEERING CORPORATION | | | FAA STC ST02755NY | A330-300 | Update of a Panasonic System eX2 In-Flight Passenger Entertainment System | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- CRI H-01 "Enhanced Airworthiness Programme for Aeroplane (...) | - FAA STC ST02755NY, dated 5th January 2010- Master Data List, ref. doc. No. 0154-06550-2200, Revision B, dated 30th December 2009- Electrical Wiring Interconnection System (EWIS) Assessment, ref. doc. No. 6654-07006-2200, Revision -, dated 1st July 2 (...) | Not applicable | 17.09.2010 |
| 10033290 | DELTA ENGINEERING CORPORATION | | | FAA STC ST01662NY | B747-400 | Modification of existing Mode S Transponders & TCAS by installation of Honeywell/ACSS P/N 7517800-10004 ATDL Transponders & 4066010-910 TCAS in accordance with Delta Engineering Master Drawing List 0107-05185-2059, Rev A, dated April 18, 2003. Validation of FAA STC ST01662NY | The Certification Basis for the original product remains applicable to this certificate/ approval. | Maintenance Manual Supplement Doc. No. 0707-05223-2059, dated February 12, 2003 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | This approval is for the replacement of the Transponder Units and TCAS. This approval does not include the transponder and TCAS system, wiring, trays, antennas, structural provisions, etc. | 11.01.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|---------------------|----------------|----------------------|--|--|--|---|--|---------------|
| 10033430 | DELTA ENGINEERING CORPORATION | | | FAA STC ST02647NY-D | ATR 42-200, ATR 42-300, ATR 42-320, ATR 42-400, ATR 42-500, ATR 72-101, ATR 72-102, ATR 72-201, ATR 72-202, ATR 72-211, ATR 72-212, ATR 72-212 A | EASA validation of FAA STC ST02647NY-D (as amended by FAA on August 31st, 2010) regarding installation of Delta Engineering Synchrois in accordance with Delta Engineering Master Data List 0101-06413-2166 Revision A dated January 13th, 2010 or later approved revisions. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: EASA Special Condition CRI H-001 Issue 02 "Enhanced Airworthiness Programme for Aeropl (...) | Delta Engineering Master Data List 0101-06413-2166 Revision A dated January 13th, 2010. Instructions for continued airworthiness as listed on the Master Data List are required with this installation/ later revisions of the above listed documents approve (...) | See those of original FAA STC ST02647NY-D. | 19.01.2011 |
| 10033596 | DELTA ENGINEERING CORPORATION | P-EASA.IM.A.S.03098 | | FAA STC ST00988NY-D | BOEING 747-400 | Installation of Matsushita Avionics Systems Corporation 2000E with AVOD1.5 Cabin Entertainment System in accordance with Delta Engineering, L.P. Master Drawing List 107-04323-250, Revision C, dated Sept. 24, 1999 or later FAA approved revisions. Validation of FAA STC ST00988NY-D. | Per FAA TCDS A20WE | Maintenance Manual Supplement for 747-400 with Matsushita Avionics Systems Corporation 2000E Cabin Entertainment System, Delta Engineering Doc. No. 707-04403-250, May 28, 2003 | Maintenance Manual Supplement for 747-400 with Matsushita Avionics Systems Corporation 2000E Cabin Entertainment System, Delta Engineering Doc. No. 707-04403-250, must be incorporated into the operator's maintenance instructions. | 30.07.2009 |
| 10036548 | DELTA ENGINEERING CORPORATION | | REV. 1 | FAA STC ST02668NY-D | A330-301, -302, -303 A330-321, -322, -323 A330-341, -342, -343 | Provision of a Panasonic Ex Phone System (Validation of FAA STC ST02668NY-D) This STC installs provisions (antenna and wiring) for Panasonic eX phone System. The Revision 1 of this STC corrects the applicability by adding two aircraft models, omitted at the initial approval. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List, Ref : 0154-07160-2234, dated 08 September 2011. Instruction for Continued Airworthiness, Ref : 0754-07163-2234 Revision A or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | This STC approval only certifies the provisions for the Panasonic eX Phone. This device activation, for which these provisions are intended, will be certified separately. | 10.11.2011 |
| 10037166 | DELTA ENGINEERING CORPORATION | | | | A330-301, A330-302, A330-303 A330-321, A330-322, A330-323 A330-341, A330-342, A330-343 | Panasonic Global Communication System - System installation This STC covers the installation of Panasonic Global Communication System (GCS) and activates the wireless portion. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List, Ref : 0154-06950-2224, revision A, dated 25 October 2011. Aircraft Flight Manual Supplement, Ref : 0654-07479-2224, dated 28 October 2011. Instruction for Continued Airworthiness, Ref : 0754-06958-2224, revision A (-, -) dated 15 May 2011 or (...) | None | 07.11.2011 |
| 10037167 | DELTA ENGINEERING CORPORATION | | REV. 1 | FAA STC ST02884NY-D | A330-201 / -202 / -203 A330-223 / -243 A330-301 / -302 / -303 A330-321 / -322 / -323 A330-341 / -342 / -343 | Panasonic Global Communication System - Antenna installation This STC installs a large antenna for the future use of Panasonic Global Communication System (GCS), on A330-300 aircraft models. Revision 1 of this STC extends the applicability to the A330-200 aircraft models. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List, Ref : 0143-06498-2193, revision D, dated 21 October 2011. A330-200 Aircraft Flight Manual Supplement, Ref : 0600-07537-2193, dated 03 October 2011. A330-300 Aircraft Flight Manual Supplement, Ref : 0600-07478-2193, dated 28 October 2011. A330 (...) | None | 16.03.2012 |
| 10037970 | DELTA ENGINEERING CORPORATION | | | FAA STC ST02888NY-D | 777-300ER | STC ST02888NY-D: Activation of Panasonic nMAP's (Wi-Fi) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 ICA on EWIS CRI F-01 WLAN | STC ST02888NY-D "Activation of Panasonic nMAP's (Wi-Fi)" Master Data List 0143-07446-2286 Rev. B dated Sep. 28, 2011 - AFM Supplement 0643-07449-2286 Rev. - dated Sep. 27, 2011 - Instructions for Continued Airworthiness 0743-07450-2286 Rev. A dated Sep. 24 (...) | Not Applicable | 10.01.2012 |
| 10038251 | DELTA ENGINEERING CORPORATION | | | FAA STC ST02890NY-D | A319 | Installation of Panasonic Global Communication System Antenna | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: SC-H01 Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS as included in TCDS EASA A.064 Issue 7 dated 13/06/2011 | FAA STC ST02890NY-D dated 08/11/2011 ODA Managed Project ST6918NY-T Supplemental Type Certificate Project Plan ODA-2268-01 Rev B dated 22/06/2011 Instructions for Continued Airworthiness - 0764-07343-2268 Rev.A dated 28/10/2011 Master Data List - 0164-0 (...) | None | 01.02.2012 |
| 10038944 | DELTA ENGINEERING CORPORATION | | | FAA STC ST02889NY-D | A330-201, -202, -203 A330-223, -243 | Panasonic Global Communication System - System installation This STC installs of Panasonic Global Communication System (GCS) and activates the wireless portion. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List, Ref : 0154-07345-2269, revision F, dated 09 February 2012. Aircraft Flight Manual Supplement, Ref : 0654-07840-2269, dated 09 October 2011 (Aircraft Configuration 1) Aircraft Flight Manual Supplement, Ref : 0654-07841-2269, dated 04 Jan (...) | None | 23.03.2012 |
| 10040102 | DELTA ENGINEERING CORPORATION | | REV. 1 | FAA STC ST02903NY-D | B747-400 | Installation of Panasonic eX2 In-Flight Entertainment system with XPhone in accordance with Delta Engineering Master Data List 0107-07609-2280 Rev. E, dated September 13, 2012 or later FAA approved revisions. Validation of FAA STC ST02903NY-D. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: Section "Title", "Ventilation", "Com", "Maintenance" and "Performance" of Amendment 25-8925.8 (...) | 0107-07608-2280 Rev E, dated September 13, 2012 Master Data List 0107-07609-2280 Rev. B, dated August 5, 2012 Instructions for Continued Airworthiness for Panasonic eX2 IFE system 0607-07662-2280 Rev. A, dated September 6, 2012 Airplane Flight Manual S (...) | The Panasonic eXPhone system must interface with the following existing systems/components which must be installed in a manner acceptable to the authorities. The below listed required existing systems are not approved by this STC. Required Supplemental (...) | 30.10.2012 |
| 10040214 | DELTA ENGINEERING CORPORATION | | | FAA STC ST02901NY-D | B747-400 | Installation of a Panasonic eX2 In-Flight Entertainment System in accordance with Delta Engineering Master Data List 0107-07459-2283 Revision B, dated May 9, 2012 or later FAA approved revisions. Validation of FAA STC ST02901NY-D. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Section "Title", "Com", "Maintenance" and "Performance" of Amendment 25-831 "Ventilation", "Com" and "Performance" of Amendment 25-8925.853, "Com" (...) | 0107-07459-2283 Rev B, dated May 9, 2012 Master Data List 0107-07464-2283 Rev. A, dated May 17, 2012 Instructions for Continued Airworthiness for Panasonic eX2 IFE system 0607-07463-2283 Rev. - Airplane Flight Manual Supplement or later revisions of the (...) | None | 18.06.2012 |
| 10041776 | DELTA ENGINEERING CORPORATION | | | FAA STC ST02714NY-D | 737-600, 737-700, 737-700C, 737-800, 737-900, 737-900ER | Installation of provisions for dual NavAero Electronic Flight Bags with NavAero Mount. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 "Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS" Issue 0 (...) | Master Data List 0106-06191-2169 ("Configuration 2") revision D dated January 15th, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | Only for installation of NavAero Mount ("Configuration 2"). The installation of ECS Mount ("Configuration 1") is not certified. | 15.10.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|----------------------|--|--|---|--|--|---------------|
| 10042191 | DELTA ENGINEERING CORPORATION | | | FAA STC ST02049NY | B747-400 | Installation of Panasonic Avionics Corporation System 3000 Passenger Entertainment System in accordance with Delta Engineering Master Drawing List 0107-05381-2073, Revision D, dated June 08, 2005 or later approved revision. Validation of FAA STC ST02049NY. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Title 14 CFR Part 25 including amendments 25-1 through 117 for 25.301, 25.303, 25.305, 25. (...) | 0107-05381-2073, Rev D, dated June 8, 2005 Master Data List, 0707-05387-2073 Rev. -, dated March 24, 2005 Instructions for Continued Airworthiness or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004 (...) | None. | 14.11.2012 |
| 10042528 | DELTA ENGINEERING CORPORATION | | REV. 2 | FAA STC ST003056CH | 737-800 | Validation of FAA STC ST003056CH: Installation of Panasonic Global Communications suite (GCS) System Provisions in accordance with Armstrong Aerospace Master Data List DC104-9041-00, Revision B, dated 22 December 2011. Revision 1: Transfer of STC 1003843 (...) | CS-25 amendment 2 for affected parts, except: - JAR 25.571 (CRI A11-05) and JAR 25.581 (change 13) as per TCDS-CRI H-01/EWIS at issue 3 dated 17 Feb 2012. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Data List DC104-9041-00, Revision B, dated 22 December 2011; - AFM Supplement DC303-9041-00, Initial release, dated 19 December 2011; - EWIS/ICA source document DC113-9041-00, Revision A, dated 28 October 2011; - Structural Repair Manual SRM to (...) | 1) The Panasonic Global Communication Suite System for which these provisions are installed and whose provisions are activated in accordance with a separate STC, EASA STC 10038436.2) The aircraft operator must perform the instructions for continued airworthiness (...) | 25.01.2013 |
| 10042791 | DELTA ENGINEERING CORPORATION | | REV. 3 | FAA ST02926NY-D | A330-201, A330-202, A330-203 A330-223, A330-243 A330-301, A330-302, A330-303 A330-321, A330-322, A330-323 A330-341, A330-342, A330-343 | Activation of a Panasonic Global Communication Suite System. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Data List 0154-07776-2293, Rev. C, dated 28/01/2013; - Aircraft Flight Manual 0654-07783-2293, Rev. A, dated 25.01.2013; Instructions for Continued Airworthiness 0754-07784-2293, Rev. A, dated 19/09/2012; or later revisions of the above list (...) | Operational approval to use T-PEDs must be obtained separately from this STC. The Panasonic GCS must interface with the following systems/components: - T-PED FAA STC ST03106CH as validated by EASA under project ref 0010015818-001 - Antenna FAA STC ST0288 (...) | 20.03.2013 |
| 10043059 | DELTA ENGINEERING CORPORATION | EASA.IM.A.S.03097 | | FAA STC ST00740NY-D | B747-400 | Installation of Matsushita Avionics Systems Corporation 2000E Cabin Entertainment System (>-<-) in accordance with Delta Engineering, L.P. Master Drawing list 156-03918-250, Revision B (>-<-) dated Dec. 22, 1998. Validation of FAA STC ST00740NY-D. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Maintenance Manual Supplement for 747-400 with Matsushita Avionics Systems Corporation 2000E Cabin Entertainment System, Delta Engineering Doc. No. 768-03925-250, Dec. 15 (>-<-) 1998. | 1. Maintenance Manual Supplement for 747-400 with Matsushita Avionics Systems Corporation 2000E Cabin Entertainment System, Delta Engineering Doc. No. 768-03925-250, Dec. 15, 1998 must be incorporated into the operator's maintenance instructions. 2. Pri (...) | 03.06.2009 |
| 10043267 | DELTA ENGINEERING CORPORATION | | | FAA STC ST02904NY-D | 777-300, 777-300ER | Installation of a Panasonic Avionics Corporation Global Communications Suite (GCS). | The Certification Basis for the original product remains applicable to this certificate/ approval. | - Master Document List 0143-07644-2282 Rev. E dtd. 6 Oct 12; - AFM Supplement 0643-07650-2282 Rev. - dtd. 3 Jul 12; - Instructions for Continued Airworthiness & Airworthiness Limitations 0743-07645-2282 Rev. C dtd. 23 Nov 12; or later revisions of the (...) | - Operational approval to use T-PEDs must be obtained separately from this STC. The Panasonic GCS must interface with the following systems/components: - T-PED FAA STC ST03048CH as validated by EASA under project ref 0010015186-001 - Antenna FA (...) | 17.01.2013 |
| 10043555 | DELTA ENGINEERING CORPORATION | | | FAA STC ST02905NY-D | B777-300/-300ER | Panasonic Global Communication System Antenna | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. ST02905NY-D dated 29 April 2012. Delta Engineering Master Data List 0143-07375-2278 Revision A dated 12 April 2012; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent (...) | None | 06.02.2013 |
| 10044839 | DELTA ENGINEERING CORPORATION | | | FAA STC ST03168NY-D | 737-600, 737-700, 737-800, 737-900. | Installation of Bruce Aerospace cabin lighting | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Delta Engineering Master Data List 0106-08443-2340 dated March 26, 2013. Instructions for continued airworthiness (ICA) and EWIS ICA 0706-08390-2340 Rev A, applicable to configuration 1 (BR9542 and BR9545) and to configuration 2 (BR9542 and BR9543) or (...) | None | 10.05.2013 |
| 10045486 | DELTA ENGINEERING CORPORATION | | | FAA STC ST02929NY-D | A330-301, A330-302, A330-303, A330-321, A330-322, A330-323, A330-341, A330-342, A330-343. | Installation of Panasonic Global Communication System in A330 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List 0154-07357-2272, Rev. -, Dated 12 Jan 2013. Aircraft Flight Manual Supplement 0654-07364-2272, Rev. -, Dated 9 Jan 2013. Instruction for Continued Airworthiness 0754-07365-2272, Rev A, Dated 17 Nov 09. Later revisions of the above listed d (...) | This Approval is linked to the approvals hereunder, which are out of the scope of this STC and must be covered by another approval. - The demonstration of compatibility of the TPEDs (Transmitting Portable Electronic Devices). - The Panasonic GCS Antenna (...) | 28.06.2013 |
| 10045753 | DELTA ENGINEERING CORPORATION | | | FAA STC ST02044NY | 777-200 | Installation of Panasonic Avionics Corporation System 3000 passenger entertainment system. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Certification Basis for this type design change is Title 14 CFR Part 25 including amendments 25-1 through 25-112 (>-<-) equ (...) | Master Drawing List, Delta Engineering document No. 0143-05393-2074, Revision E, dated 15 April 2005. Instruction for Continued Airworthiness, Delta Engineering document No. 0743-05399-2074, Revision A, dated 3 January 2005 or later revisions of the above (...) | FAA STC ST01493SE, appropriately approved in accordance with Technical Implementation Procedures of EU/USA Bilateral Agreement, is required to be installed as a prerequisite for accomplishment of this STC. Prior to installation of this design change it mu (...) | 16.07.2013 |
| 10045841 | DELTA ENGINEERING CORPORATION | | | FAA STC ST03172NY | 747-400 | Installation of Bruce Aerospace Emergency Path Lighting System. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ODA-2350-01, Revision A, 3 May 2013 - Project Plan 6607-08432-2350, Revision -, 10 July 2013 - Applicant Statement of Compliance 0107-08433-2350, Revision A, 10 July 2013 - Master Data List 0707-08434-2350, Revision A, 03 July 2013 - Instructions for Co (...) | This STC is a prerequisite of the KLM STC with project number 0010018346-001. Prior to installation of this design change must be determined that the interrelationship between this design change and any other previously installed design change and/or (...) | 24.07.2013 |
| 10015942 | DENEL OPTRONICS | EASA.IM.A.S.02084 | | - | CESSNA 208 AND 208B | Installation of Airborne Observation System | | -- | 1. Max operating speed 155 kts IAS2. Observation system must be switched OFF during take-off and landing. 3. Persons on board during flight: limited to flight crew and operator(s) of observation system | 28.07.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|--------------------------------|---|---|--|--|---|---------------|
| 10015483 | DEVORE AVIATION CORPORATION OF | EASA.IM.A.S.01243 | | FAA STC ST09191AC | BOEING 737-200 BOEING 737-200C BOEING 737-300 BOEING 737-400 BOEING 737-500 | Installation of DeVore Wing Forward Position Light Assembly according to FAA STC ST09191AC | | FAA STC No ST09191AC, Date reissued 5 May 2000 DeVore Top Drawing No 73700002, Rev E dated 20th August 2003 | 1. According to Limitations Section of FAA STC ST09191AC2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.10.2005 |
| 10015937 | DEVORE AVIATION CORPORATION OF | EASA.IM.A.S.02078 | REV 1 | FAA STC SA7473SW | BOEING 757 | Installation of tail floodlights in accordance with DeVore Drawing No. 75700001, Rev B, dated September 14, 1988, or later FAA approved revision. | | Modification FAA STC No. SA7473SW | | 02.04.2007 |
| 10015947 | DEVORE AVIATION CORPORATION OF | EASA.IM.A.S.02089 | | SA998EA | BOMBARDIER DHC-6 SERIES 100 BOMBARDIER DHC-6 SERIES 200 BOMBARDIER DHC-6 SERIES 300 | Installation of Engine Fuel Drain Recirculation System | | FAA STC SA998EA DeVore Aviation Corporation Installation Instruction Manual F-200 Revision B dated April 21, 1975 or later approved revision | | 18.10.2006 |
| 10016743 | DEVORE AVIATION CORPORATION OF | EASA.IM.R.S.01069 | | FAA STC SH4694SW | S-76A/B/C | Installation of Under Fuselage forward facing recognition lights and rotor disc lights | | Top Drawing 07600002 Revision C dated August 16, 2004 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 31.05.2006 |
| 10016744 | DEVORE AVIATION CORPORATION OF | EASA.IM.R.S.01070 | | FAA STC SH5867SW | AS 365N/N1/N2, EC 155B/B1 | Installation of Tail Floodlights and Associated Technical Documentation DeVore Drawing List 36500001 Revision J dated March 16 1999 DCN F-1 dated February 8 1989 & DCN L-1 dated October 22 2001 or later approved revision | | DeVore Drawing List 36500001 Revision J dated March 16 1999 DCN F-1 dated February 8 1989 & DCN L-1 dated October 22 2001 or later approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 21.06.2005 |
| 10016745 | DEVORE AVIATION CORPORATION OF | EASA.IM.R.S.01071 | | FAA STC SH5866SW | AS 365N/N1/N2, EC 155B/B1 | Installation of Forward Facing Recognition Lights | | DeVore Drawing List 36500002 Revision C dated July 2 2001 or later approved revision. DeVore Aviation Corporation RFM Supplement dated June 10 2002 or later approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 21.06.2005 |
| 10016746 | DEVORE AVIATION CORPORATION OF | EASA.IM.R.S.01072 | | FAA STC SH7579SW | AS 365N1/N2/N3, EC 155B/B1 | Installation of a rotor disc light and associated DeVore Aviation Corporation Top Drawing-Rotor Disc Light Inspection; Aerospatiale/Dauphin Drawing 36500003 revision E dated August 21 2001 or later approved revision DeVore Aviation Corporation RFM S (...) | | DeVore Aviation Corporation Top Drawing-Rotor Disc Light Inspection/Aerospatiale/Dauphin Drawing 36500003 revision E dated August 21 2001 or later approved revision/DeVore Aviation Corporation RFM Supplement dated June 10 2002 or later approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.06.2005 |
| 10017026 | DEVORE AVIATION CORPORATION OF | EASA.IM.R.S.01445 | | FAA STC SH3754SW | | Installation of DeVore forward facing recognition lights | | DeVore Top Drawing 10900002 Rev. C dated 17 May 2004 or later FAA approved revision DeVore DR-317 | | 01.07.2008 |
| 10028007 | DEVORE AVIATION CORPORATION OF | | | FAA STC SR09557RC | R44II | Installation of Tail/Rotor, Forward Facing Recognition and Fuselage Floodlights in accordance with DeVore Aviation Corporation Drawing Number 44000001. This installation is based on FAA STC SR09557RC. | FAR 27 as per EASA TCDS The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Flight Manual Supplement DR-708 Rev. IR, dated July 29, 2009 DeVore Aviation Corporation Drawing Number 44000001, Revision A, dated July 20, 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004 (...) | None | 24.11.2009 |
| 10016995 | DIAMOND J, INC. | EASA.IM.R.S.01404 | | FAA STC NO. SR00031LB DATED MA | BELL 206B BELL 206L BELL 206L-1 BELL 206L-3 BELL 206L-4 | Installation of Analog/Digital Turbine Outlet Temperature (TOT) Indicator with Exceedance logging Capability. This STC has been classified as "Basic" by FAA and has been approved under the PTV. | | See attached sheet. | | 05.03.2008 |
| 10044868 | DIAMOND UPGRADES, L.L.C. | | | FAA STC SR01180LA | 412 | PART 1: INSTALLATION OF PRATT-<-&->WHITNEY CANADA PT6T-3DF TWIN-PACPOWER PLANT WITHOUT INCREASED PERFORMANCE CAPABILITY. PART 2: INSTALLATION OF PRATT-<-&->WHITNEY CANADA PT6T-3DF TWIN-PACPOWER PLANT AND INSTRUMENT CHANGE WITH INCREASED PERFORMANCE (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement for Bell Model 412 AU-412010423 REV.2 Dated March 11, 2004 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) (...) | -.PART 1: Refer to the basic Bell RFM. Same performance and limitations as the previously installed PT6T-3B, -3BE, -3BF, -3BG engine. -.PART 2: Refer to RFM Supplement AU-412010423 for increased OE performance of the PT6T-3DF with the appropriate en (...) | 13.05.2013 |
| 10015149 | DIEHL SERVICE MODULES GmbH | EASA.A.S.03509 | REV 1 | | BOEING 777-200 | Installation of Galleys and Closets on BAW B777-236, Kit P/N 190331-51 | | Certification Program and Compliance Checklist CP 190331 dated February 27th, 2008 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The i (...) | 24.11.2008 |
| 10014433 | DINAN AVIATION | EASA.A.S.02481 | | | PIPER PA-31T | Modification SIBA MD01_2006 Replacement of one ATC mode C by Mode S and replacement of an old DME with adding an ATC antenna on the top of frame | | None | This STC is limited to: S/N 31T-7920094 F-GPBF | 31.08.2006 |
| 10030311 | DINAN AVIATION | | | | PA28 R 201 T | M05-2009 Rev. 0 modernisation du système de communication. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Supplement au manuel de vol PA28 Rev0 édition Octobre 2009 or later revisions of the above listed documents approved by EASA. | None. | 10.06.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|-------------------|----------------|----------------------|--|--|--|---|---|---------------|
| 10042971 | DIVERSIFIED AVIATION SERVICES INC. | EASA.IM.A.S.01171 | | | DC.9-31/-32/-34 | Installation of a photo luminiscent floor proximity emergency escape path marking system. | | FAA AFMS DAS Report No. J17-11IR dated 4 October 1999. Dispatch Operating Procedures DAS report No. J17-13IR dated 4 October 1999, or later approved revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.07.2005 |
| 10039614 | DONICA INTERNATIONAL LIMITED | | REV. 7 | | SEE EASA APPROVED MODEL LIST | Installation of Donica Digital Media Reproducer | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Modification DN-DMR-23-0000 Issue 6, dated 8th July 2013. Instructions for Continued Airworthiness V.03, Feb 2013. Service Bulletins DN-MD90-23-000-01 V.02 Installation of DMR onto MD90-30 (Eva Air) - Raised in issue DN-737-23-0000-02 V.01 Installation o (...) | Other: Amendment to the STC is required in order to extend the applicability of this STC to the following types, A300-600/A310 (TCDSGAC France 145), A318 (EASA.A.064), A330 (EASA.A.004), A340 (EASA.A.015), B747 (EASA.IM.A.196), B757 (EASA.A.205), B77 (...) | 16.07.2013 |
| 10017285 | DRF STIFTUNG LUFTRRETTUNG | EASA.R.S.01533 | | | EUROCOPTER DEUTSCHLAND BK117 B | Title: Installation of a L-3 SSCVR Cockpit Voice Recorder. Description: Optional Installation of a Cockpit Voice Recorder. The installation is in accordance with DRF change no. 135, MDL MD-117-31-32-00001. | | Flight Manual Supplement: # BK117 B-2 Flight Manual Supplement FMS 10-DRF-001 Rev. A. Installation Instructions / Instructions for Continued Airworthiness: # Installation Manual IM-117-31-32-00001# Instructions for Continued Airworthiness CA-117-31-32-00001 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 04.05.2009 |
| 10029942 | DRF STIFTUNG LUFTRRETTUNG | | | | MBB-BK 117 B-2, MBB-BK 117 C-1 | Installation of Garmin GNS 530 FMS COM/NAV System and TCAS-1 Sky 899TCAS | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS29 Amdt. 2 The certified noise and/ or emissions levels of the original product are (...) | Flight Manual Supplement SKY-899 TCAS I FMS-10-DRF-003 Rev. A Flight Manual Supplement GNS-530W, FMS-10-DRF-004 Rev. A or later revisions of the above listed documents approved by EASA - Master Document List MD-117-34-60-00001 Rev. A - ICA CA-117-34-60-00012 Rev. A - ICA CA-117-34-60-00001 Rev. A | Not Applicable | 04.05.2010 |
| 10031819 | DRF STIFTUNG LUFTRRETTUNG | | | | MBB-BK117B-2, MBB-BK117 C-1 | Installation of second directional gyro system installation of a second directional gyro system for IFR operation based on DRF Major Change Project 240 and defined in MD-117-34-20-00001. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CS 29 Amdt. 2 for the change. The requirements for environmental protection and the assoc (...) | Flight Manual Supplement 10-DRF-05, Rev. 1. Or later revisions of the above listed documents approved by EASA - Installation Manual IM-117-34-20-00001, Rev. 00 - Instr. f. Continued Airworthiness CA-117-34-20-00001, Rev. A - Master Document List MD-117- (...) | None | 16.09.2010 |
| 10033429 | DRF STIFTUNG LUFTRRETTUNG | P-EASA.R.S.01563 | | | BK117 C-1 | Title: Installation of Modular Airborne Data Recording/Acquisition System (MADRAS) Description: Optional Installation of a Cockpit Voice and Flight Data Recorder System. The installation is in accordance with DRF change no. 111 (-> -<)- MOL MD-117-31-32-00003 Rev. A or later approved revision. | CS-29 Amdt. 2 | - Installation in accordance with DRF change no. 111, MDL MD-117-31-32-00003 Rev. A Flight Manual/Flight Manual Supplement - BK117 C-1 Flight Manual Supplement FMS 10-DRF-002 Rev. A Technical Instructions - Installation Manual IM-117-31-32-00002 Rev. A- (...) | None | 23.06.2009 |
| 10041970 | DRF STIFTUNG LUFTRRETTUNG | | | | MB BK 117 C-2 | Installation of a TETRA Digital Radio System into a NVIS certified Cockpit. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI on NVIS dated 11.09.2012 The requirements for environmental protection and the asso (...) | ...FMA 11-5-1 APPENDIX FOR TETRA Communication System rev. A dated 27.08.2012 or later revisions of the above listed documents approved by EASA - Instructions for Continued Airworthiness Doc. N°: CA-117-23-10-00025 Revision: 00 or later approved revisions of the above listed documents. | Required premodification: EASA approval EASA 10034057 (EC D major change E-1807). This rotorcraft employs Night Vision Imaging System lighting, commonly named NVIS lighting, that has been certified to ensure the aircraft is compatible with night vision go (...) | 29.10.2012 |
| 10016181 | DRIESEN AIRCRAFT INTERIOR SYSTEMS | EASA.IM.A.S.02410 | | FAA STC ST00826LA | AIRBUS A340-300 | Installation of Lower Deck Mobile Crewrest. | | # FAA STC ST00826LA dated 15 April 2003# Master Data List, Ref. MDL-101682, Rev. C, dated April 1st 2002.# Component Maintenance Manual with Illustrated Part List, Ref. 25-70 03, dated 30th September 2001# or later revisions of the above (...) | 1. The limitations and conditions as referenced in FAA STC ST00826LA apply. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modif (...) | 22.08.2007 |
| 10016292 | DRIESEN AIRCRAFT INTERIOR SYSTEMS | EASA.IM.A.S.02588 | REV. 1 | FAA STC ST00843LA | BOEING 757-200 SERIES | Galley installation. | | Driessen Master Data List MDL-101152 Revision C, 23/06/2000. | None. | 18.02.2010 |
| 10016552 | DRIESEN AIRCRAFT INTERIOR SYSTEMS | EASA.IM.A.S.02932 | | | BOEING 747-200B BOEING 747-200C | Installation of a TBO Container. | | Driessen Aircraft Interior Systems Data List MDL-101969, Revision "C" dated April 14, 1998. | The installation should not be incorporated in any aircraft unless it is determined that the relationship between this installation and any previously approved configuration will not introduce any adverse effect upon the Airworthiness of the Aircraft. (...) | 04.07.2008 |
| 10037862 | DUGAN KINETICS, L.L.C. | | | FAA STC ST02127SE | DC-9-81/-82/-83/-87, MD-88 | Thrust Reverser/Installation of ejector thrust reverser on Pratt & Whitney Turbofan JT8D-217, -217A, 217C or -219 Engines as per FAA STC ST02127SE dated 08 Apr. 2011. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Dugan Kinetics Master Drawing List (MDL) No. 115-100, Revision A, dated 14 Sept. 2010 Dugan Kinetics Maintenance Program Supplement (MPS) Manual - Appendix B No. MPS115-307, Revision B, dated 25 Feb 2011 Instruction for Continued Airworthiness (ICA) Manua (...) | Refer to FAA STC ST02127SE. | 03.01.2012 |
| 10016061 | DUGANAIR TECHNOLOGIES, IC | EASA.IM.A.S.02239 | | FAA STC ST01182SE | B737-200, 737-300, 737-400 B737-500 | Installation of Duganair Security Flight Deck Door as defined in FAA STC ST01182SE | | - Duganair Technologies Inc., Index List, QS-00001-3A, Revision B, dated August 27, 2003 or later approved revisions;- FAA Approved Airplane Flight Manual Supplement No. QWC737101-4, Rev. 1, dated December 16, 2004 or later approved revision is required as part of this installation. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 15.11.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------|-------------------|----------------|-------------------------|--|--|---------------------|---|--|---------------|
| 10015358 | DUNCAN AVIATION, INC. | EASA.IM.A.S.00657 | | FAA STC ST00515W1 | LEARJET 35 LEARJET 35A LEARJET 36 LEARJET 36A | Installation of an Allied Signal CAS 67A Traffic Alert & Collision Avoidance (TCAS II). (STC ST00515W1) | | - Duncan Aviation Inc. Document List No. 10048800 Revision B dated March 20, 1997, or later approved revisions.- Duncan Aviation Inc.-AFM Supplement Document No. 40169100 dated August 9, 2004, or later approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 04.04.2005 |
| 10015417 | DUNCAN AVIATION, INC. | EASA.IM.A.S.01139 | | FAA STC ST00546W1 | LEARJET 60 | Installation of a Universal Avionics Inc. UNS-1B FMS (Dual). | | -FAA STC ST00546W1.-Duncan Aviation Inc. Document List No10049800, May 27, 1997, or later FAA-approved revisions.-Duncan Aviation Inc; Document No 40173300, FAA-approved AFMS, June 13, 1997, or later FAA-approved revisions. | 1. For certification basis refer to FAA TCDS A10CE.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.04.2005 |
| 10015488 | DUNCAN AVIATION, INC. | EASA.IM.A.S.01249 | | FAA STC ST00452W1 | CESSNA 650 CITATION III CESSNA 650 CITATION IV CESSNA 650 CITATION VII | Installation of the AlliedSignal CAS 67A TCAS II System | | Duncan Aviation, Inc., Document List No 10047000, Revision A, dated May 20 1996Duncan Aviator, Inc., Document List No 40160000, FAA approved Airplane Flight Manual Supplement, dated May 31, 1996 Revision A, dated May 20, 1996orDocument No40165900, (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 13.03.2006 |
| 10015505 | DUNCAN AVIATION, INC. | EASA.IM.A.S.01268 | | ST00315W1 | LEARJET 60 | Installation of Collins TCAS-94 I.a.w. Dok List 10045000 | | AFM 40150000 | | 07.02.2007 |
| 10015640 | DUNCAN AVIATION, INC. | EASA.IM.A.S.01603 | | FAA STC ST0069W1 | BOMBARDIER CL-600-1A11 (CL-600 BOMBARDIER CL-600-2A12 (CL-601 BOMBARDIER CL-600-2B16 | Installation of Honeywell TCZ-910 TCAS II & XS-852 Mode S Transponder System | | 1) Duncan Aviation Inc. Document List No. 10040700 RevisionG dated December 3, 2002) Duncan Avation Inc. AFM Supplement Document Nos 40134800 Revision B & 40141800 Revision B; or later FAA approved revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 25.10.2005 |
| 10015742 | DUNCAN AVIATION, INC. | EASA.IM.A.S.01766 | | FAA STC SA00894W1 | CESSNA 525 | Installation of Honeywell CAS 67A TCAS II System on Cessna 525. FAA Approval of STC Nr. SA00894W1 08 September 2000. | | | The call outs and their priority are to be confirmed as part of each installation. | 26.01.2006 |
| 10015826 | DUNCAN AVIATION, INC. | EASA.IM.A.S.01904 | | ST00932W1 | LEARJET 31 LEARJET 35 LEARJET 35A LEARJET 36 LEARJET 36A | Installation of Universal Avionics TAWS | | Duncan Aviation Document List 10055100 Revision A dated December 14, 2000 or later approved revision.Duncan Aviation AFM Supplement 40010400 dated December 18, 2000 or later approved revision. | None | 11.05.2006 |
| 10015844 | DUNCAN AVIATION, INC. | EASA.IM.A.S.01941 | | FAA STC SA2405CE-D | GULFSTREAM 1125 WESATRD ASTRA GULFSTREAM ASTRA/SPX | Installation of Garrett Model GTCP36-150(W) Auxiliary Power Unit used as nonessential Bleed Air source for on board electrical and environmental systems. | | 1125 Astra:Duncan Aviation Drawing list No AST 001, dated 15 Sept 88 or later FAA approved revision. Duncan Aviation Inc, FAA approved AFM supplement, dated 16 Sept 88Astra SPX:Duncan Aviation Drawing list No AST 001, dated 5 January 96 or later FAA (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 25.10.2006 |
| 10015917 | DUNCAN AVIATION, INC. | EASA.IM.A.S.02049 | | FAA STC ST00865CH-D | BAE 125-800 SERIES HAWKER 800 HAWKER 800 XP | Installation of Honeywell MK VII Enhanced Ground Proximity Warning System (EGPWS) to software version -Z26-226 and Optional RAAS | | # Installation in accordance with Duncan Aviation MasterDrawing List KA9807-01R, Revision L, dated January 25, 2005or later approved revisions.# Aircraft to be operated in accordance with Aircraft Flight Manual Supplement KA9807-04R, Revision F dated December 9, 2004, or later approved revisi | 1. Only Software standard -Z26-226 or later is approved for installation.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modifi (...) | 10.09.2007 |
| 10015938 | DUNCAN AVIATION, INC. | EASA.IM.A.S.02079 | | ST01157WI-D | LEARJET 35A LEARJET 36A | Installation of L-3 Communications F1000, SSFDRFAA STC ST01157WI-D | | Certification Plan 40415300, rev. - (Model 35A)Certification Plan 40414400, rev. C (Model 35A)Certification Plan 060501017, rev. A (Model 36A)AFM Supplement 40026900, rev. A (Model 35A, 36A) | | 30.10.2006 |
| 10015972 | DUNCAN AVIATION, INC. | EASA.IM.A.S.02123 | | FAA STC ST01159WI-D | LEARJET 31 LEARJET 35A LEARJET 36A | Installation of Honeywell Mk-VIII EGPWS.FAA STC ST01159WI-D | | Certification Plan 060421001, rev.-AFM Supplement 40026700,rev. A. | None. | 17.01.2007 |
| 10015987 | DUNCAN AVIATION, INC. | EASA.IM.A.S.02141 | | ST02092CH-D | HAWKER 800 XP | Installation of a forward cabin entry door and aft cabin reconfiguration | | Installation to be in accordance with Duncan Aviation Master Drawing list 060502016 , Rev A, dated, July 7, 2006 or later approved revision.Continued Airworthiness to be in accordance with Duncan Aviation Instructions for Continued Airworthiness ref (...) | Applicable to Aircraft Serial Number 258645 only | 20.09.2006 |
| 10016104 | DUNCAN AVIATION, INC. | EASA.IM.A.S.02294 | | FAA STC NO. ST2359CH-D | BAE 125 SERIES 800A | Modification of the existing Collins TDR-94D Transponders and Collins RTU-870 Radio Tuning Units to Enhanced Surveillance capable equipment and aircraft interface requirements. | | # Modification to be in accordance with Duncan Aviation Master Drawing List Document No. 061103015, Revision A, dated December 19, 2006, or later approved revision.# Operation to be in accordance with Duncan Flight Manual Supplement Document N (...) | 1. This STC is only approved where ARINC 429 databus or similar standard is used as an input to the transponders, itdoes not approve the use of Gilham or Grey code as the input to the Mode S transponders. As the applicant has not demonstrated tha (...) | 20.08.2007 |
| 10016105 | DUNCAN AVIATION, INC. | EASA.IM.A.S.02295 | | FAA STC NO. ST02370CH-D | BAE 125 SERIES 800A HAWKER 800 HAWKER 800XP | Upgrade of the existing dual Universal Avionics Systems Corporation UNS-1D Flight Management System to UNS-1D+ Flight Management System. | | Duncan Aviation Master Drawing List 061108001, Revision A, dated December 22, 2006 or later FAA approved revision. | EASA approved Airplane Flight Manual Supplement, 061108004, Revision 0, dated November 10, 2006. EASA approved January 10, 2007 or later EASA approved revision is required.Compatibility of this design change with previously approved modifications mus (...) | 31.05.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|---------------------|----------------|--------------------------|---|---|---|--|---|---------------|
| 10016110 | DUNCAN AVIATION, INC. | EASA.IM.A.S.02300 | | US STC NO. ST02371CH-D | HAWKER 800 XP HAWKER 800, 800A | Installation of AirCell Iridium ST-3100 single channel SATCOM system. | | Duncan Aviation Master Drawing List 041008020, Revision A, dated December 22, 2006 or later FAA approved revision. | Compatibility of this design change with previously approved modifications must be determined by the installer. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. | 31.05.2007 |
| 10016149 | DUNCAN AVIATION, INC. | EASA.IM.A.S.02351 | | FAA STC NO. ST02091 CH-D | BAE 125 SERIES 800A HAWKER 800 HAWKER 800 XP | Replacement of Existing 3, 4 and 5 auxiliary batteries with Securaplane XL245 lead acid batteries | | Duncan Aviation Master Drawing List 060412006, Revision C, dated December 20, 2006 or later approved revision | 1. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. | 30.04.2007 |
| 10016235 | DUNCAN AVIATION, INC. | EASA.IM.A.S.02499 | | FAA STC ST00858W1 | LEARJET 35 LEARJET 35A LEARJET 36 LEARJET 36A | Installation of Collins TCAS-94 TCAS II System. | | AFM Supplement 40003000, ref | This installation is not to be used in airspace where Elementary Surveillance (ELS) or enhanced surveillance (EHS) is required. | 19.12.2007 |
| 10016392 | DUNCAN AVIATION, INC. | EASA.IM.A.S.02721 | | FAA STC ST00914W1-D | LOCKHEED 1011-385-1 | Installation of Dual IS&S Air Data/ Altimeter System along with two inch standby pneumatic altimeter (FAA STC ST00914W1-D). | | Master Document List 10055000, Rev. P. Supplement to Aircraft Flight Manual (AFM) DOC. 40008200, Rev. D. Supplement to AFM(S/N 5086 only) Doc 050822006, Rev. 0. Instructions for Continuous Airworthiness Doc 44505300, Rev. A. | approval may be extended to aircraft models Lockheed 1329-23, previously modified law FAA STC SA2326SW, when this STC has been EASA validated. Lockheed 1329-25 Group Approval for RVSM requirements can be shown in conjunction with FAA STC ST010625E (Ae (...)) | 26.05.2008 |
| 10016393 | DUNCAN AVIATION, INC. | EASA.IM.A.S.02722 | | FAA STC ST01059W1 | LOCKHEED 1011-385-1 | Installation of Honeywell MK VII Enhanced Ground Proximity Warning System (EGPWS). FAA STC ST01059W1. | | Master Document List 10057500, Rev. C. Supplement to Aircraft Flight Manual (AFM) Doc. 40018500, Rev. A. Instructions for Continuous Airworthiness Doc. 44509200. | This Design Approval may be extended to Airplanes Model Lockheed 13-29-23, previously modified law FAA STC SA2326SW, when validated by EASA. Prior to installation of this modification the installer must determine that the interrelationship between the (...) | 13.10.2008 |
| 10026867 | DUNCAN AVIATION, INC. | P-EASA.IM.A.S.03211 | | FAA STC ST00703W1 | CESSNA 650 | Installation of an AlliedSignal Enhanced Ground Proximity Warning System (EGPWS) | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC ST00703W1 Document List 10051900 Revision K from 12 June 1998 Airplane Flight Manual Supplement (AFMS) 40189700 Revision D from 16 Sept. 1998 Instructions for Continued Airworthiness (ICA) 44500400 from 12 June 1998 or later revisions of the above (...) | According to associated technical documentation, FAA STC, AFMS and ICA | 18.08.2009 |
| 10028927 | DUNCAN AVIATION, INC. | P-EASA.IM.A.S.03046 | | FAA STC ST01409W1-D | MYSTERE FALCON MF900 | EASA Validation of FAA STC ST01409W1-D Installation of an L3 Communications, GH-3100 Electronic Standby Instrument System (ESIS) in accordance with Duncan Aviation Inc Master Document List No 06062208, Revision C, dated February 19th, 2008 or later revision (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Duncan Aviation Master Document List No 06062208, Revision C, dated February 19th, 2008 - Duncan Aviation Aeroplane Flight Manual Supplement, No. 060622020, Revision A, originally approved October 10th, 2006 or later revisions of the above listed documents (...) | Validation limited to Basic Mystere Falcon MF900 aircraft model | 19.02.2010 |
| 10040935 | DUNCAN AVIATION, INC. | | | FAA STC ST00600CH-D | MF 900 & F900EX | EASA validation of FAA STC ST00600CH-D issue dated 12 Sept 2008 Installation of Honeywell Enhanced Ground Proximity Warning System with Windshear Detection and optional Terrain Display (SW level -218-218) and optional Runway Advisory Awareness System (RAAS) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Duncan Aviation Master Drawing List KA9611-23R, revision J dated July 24, 2012 FAA approved Airplane Flight Manual Supplement, KA9611-27R, Revision E, dated April 2, 2004, FAA approved May 18, 2004 or later revisions of the above listed documents approved (...) | As per FAA STC ST00600CH-D issue dated 12 Sept 2008 | 06.08.2012 |
| 10042945 | DUNCAN AVIATION, INC. | EASA.IM.A.S.01072 | | | CESSNA 750 | Installation of an AirCell Iridium ST-3100 Single Channel SATCOM | | -Duncan Aviation Drawing List DA0211-01R Revision A dated December 27 2002 or later approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.03.2005 |
| 10044932 | DUNCAN AVIATION, INC. | | | FAA STC ST10939SC | 560XL | Installation of a Norwegian Special Mission (NSM) Flight Inspection System UNIFIS 3000 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List 768-003-MDL_560XL Revision F from 15 April 2013 Airplane Flight Manual Supplement 768-023-AFMS_560XL Initial Revision from 03 March 2009 Instructions for Continued Airworthiness 768-025-ICA_560XL Revision E from 15 April 2013 or later (...) | Applicability limited to Variant XLS (SER 560-5501 to 560-5830) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will (...) | 21.05.2013 |
| 10028837 | DUNLOP AIRCRAFT TYRES LIMITED | P-EASA.A.S.03864 | REV. 1 | | EMBRAER ERJ 190-100 ECJ EMBRAER ERJ 190-100IGW EMBRAER ERJ 190-100LR EMBRAER ERJ 190-100STD EMBRAER ERJ 190-200IGW EMBRAER ERJ 190-200LR EMBRAER ERJ 190-200STD | | | | | 29.06.2009 |
| 10038094 | DUNLOP AIRCRAFT TYRES LIMITED | 0010014209-001 | | TCCA STC SA02-29 | CL-600-2B19 | Installation, Dunlop Aircraft Tyres Ltd P/N DR27321T Main and DR15112T Nose Gear Tyres This STC is issued in order to reflect the transfer of ownership from ACS-NAI Ltd to Dunlop Aircraft Tyres Ltd. This STC has previously been validated in the followin (...) | EASA TCDS IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are (...) | -Aero Consulting Services Modification Data Summary Number D08500 Revision A or later approved revision -Dunlop Aircraft Tyres Ltd. Document Number DM1172 General Servicing Instructions -Instructions for Continued Airworthiness - KTS Aeronautical Inc.D (...) | Different nose landing gear part number tires may not be intermixed on the same axle. | 18.01.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|----------------|----------------|----------------------|--|--|---|---|--|---------------|
| 10039387 | DUNLOP AIRCRAFT TYRES LIMITED | | | | ATR 42-200/-300/-320/400/-500 | Installation of Dunlop Aircraft Tyres on ATR 42 series The STC installs Dunlop Aircraft Tyres Ltd P/N DZ3012T (Main LandingGear) and P/N DR15520T (Nose Landing Gear). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List, ref. KT509009MDL, dated 11 April 2012 Instruction for Continued Airworthiness, ref. KT509009ICA, dated 11 April 2012 later revisions of the above listed documents approved by EASA | None | 02.05.2012 |
| 10041486 | DUNLOP AIRCRAFT TYRES LIMITED | | | | ERJ 170-100 LR, ERJ 170-100 STD, ERJ 170-200 LR, ERJ 170-200 STD. | Dunlop Aircraft Tyres Limited P/N DZ32816T Radial Nose Landing Gear installation on ERJ 170. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List 170 PR5 MDL Rev IR dated 11 September 2012 later revisions of the above listed documents approved by EASA. | It is not allowable to mix different tyre types (radial and bias) in the nose landing gear leg. | 20.09.2012 |
| 10045282 | DUNLOP AIRCRAFT TYRES LIMITED | | | | ATR 72-101/-102/-201/-202 ATR 72-211/-212/-212A | Installation of Dunlop Aircraft Tyres on ATR 72 series The STC installs Dunlop Aircraft Tyres Ltd P/N DZ30730T (Main LandingGear) and P/N DR15520T (Nose Landing Gear). | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List, ref. DATL PR3 MDL Rev IR, dated 20 May 2013 Instruction for Continued Airworthiness, ref. DATL PR3ICA rev IR, dated 20 May 2013 later revisions of the above listed documents approved by EASA | None | 12.06.2013 |
| 10030818 | DYN AVIATION | | | | DR300/108, DR300/180R, DR300/120, DR400/120, DR300/140, DR300/125, DR340, DR315, DR360, DR380, DR400/100, DR400RP, DR400NGL, DR400/120A, DR400/160D, DR400/120D, DR400/180S, DR400/125, DR400/140, DR400/125L, DR400/140B, DR400/160, DR400/180, DR400/180R, DR400/2+2, DR400/200R, DR400/500 | MD09F0001 050210 - New flaps and control system | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | SI10A00001 - Instructions de maintenance later revisions of the above listed documents approved by EASA | None | 08.07.2010 |
| 10031522 | E & B HELICOPTERS LTD. | | | TCCA STC SH07-38 | 407 | Audio Alert System | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FMS703.91 revision 2 dated 24 August 2010 Flight Manual Supplement DCL703 revision 3 dated 21 September 2007 - Document Control list later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or (...)) | None | 27.08.2010 |
| 10013998 | E A D AEROSPACE | EASA.A.S.01035 | | | BEECH 1900C, AIRLINER BEECH 1900D | ELS/EHS mode S installation on B1900 series. | | -Top Master Drawing List 00886-JAA-MDL rev. D or later approved revision. -Maintenance planning document supplement DC112-0886-00A or later approved revision. -Minimum Equipment List Supplement DC109-0886-00A or later approved revision. -Airplane Fligh (...) | The Airplane Flight Manual Supplement DC107-0886-00B or later approved revisions must be used with this modification. | 12.04.2005 |
| 10013999 | E A D AEROSPACE | EASA.A.S.01036 | | | BOEING 737 | Installation of a Honeywell EGPWS Mark V. | | EAD Master Drawing List Reference 01024-EAS-A-MDL revision E.AFM Supplement Reference 01024-EAS-AFM-01 Revision D. | | 14.06.2005 |
| 10014000 | E A D AEROSPACE | EASA.A.S.01037 | | | BEECH 1900 | Installation of a Traffic Alert and Collision Avoidance System, model Honeywell CAS-67A TCASII installation on B1900 Series. | | EAD Master Drawing List Document No. 00887-JAA-MDL, Revision F dated 25 May 2005, or later EASA approved revisions. Maintenance Planning Document Supplement DC112-0887-00A or later Revisions. | All wiring required for TCAS II activation must be installed. ATC Mode S Transponders must be installed. | 18.07.2005 |
| 10014124 | E A D AEROSPACE | EASA.A.S.01234 | | | AIRBUS A300-600 AIRBUS A310 | Modification of ELT ADT406 AF on A300/A310 Installation of ELT ADT 406 AF | | -Modification 01029-MDL 01029-EAS-TMD-AFM Supplement 01029-EAS-A-AFM-01 Revision A-MPD Supplement 01029-EAS-A-MPD-01 Revision A | | 28.07.2005 |
| 10014314 | E A D AEROSPACE | EASA.A.S.01562 | REV. 1 | | FAIRCHILD SWEARINGEN SA 227 SE | Installation of TCAS II on SA 227 series aircraft TCAS II installation in accordance with EAD Supplemental Type Design Definition Document 01043-EAS-A-STD-01 rev. A # Removal of existing TCAS and associated tray in forward avionics # Installation of (...) | | Aircraft Flight Manual Supplement : Document 01043-EAS-A-AFM-01 Rev. C Maintenance Planning Document Supplement : Document 01043-EAS-A-MPD-01 Rev. B | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 03.04.2006 |
| 10014315 | E A D AEROSPACE | EASA.A.S.01563 | | | AEROSPATIALE ATR 42-300 SERIES | Installation of a calibration antenna in accordance with 01050-EAS-A-MDL. | | Maintenance Planning Document Supplement ref. 01050-EASA-A-MPD-01. | | 21.04.2006 |
| 10014749 | E.I.S. AIRCRAFT GmbH | EASA.A.S.02934 | | | CESSNA 525 CESSNA 525A | Avionic Upgrade TO DUAL GTX 330D and installation of SAT-COM Antenna. | | 52C21000-91501, Rev.01, EASA approved AFM-Supplement "GARMIN GTX 330D Mode S Transponder" 52C21001-91501, Rev. 02, EASA approved AFM-Supplement "SAT Antenna Iridium" | | 23.01.2007 |
| 10014931 | E.I.S. AIRCRAFT GmbH | EASA.A.S.03185 | | | BEECH C90 A BEECH C90KING AIR | Avionics Upgrade to Garmin GTX330D diversity Mode S Transponder. | | Installation must be in accordance with EIS Aircraft GmbH Master Document List MDL 90C21002 Issue 06, dated 20.08.07 or later Approved revisions. Operation must be in accordance with EIS Aircraft GmbH Flight Manual Supplement 90C21002-91501 Issue 02, d (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The i (...) | 21.08.2007 |
| 10015148 | E.I.S. AIRCRAFT GmbH | EASA.A.S.03508 | | | BOEING 737-700 | Cabin reconfiguration. | | N/A | Applicable for B737-700 MSN 28609. | 15.04.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------|-------------------|----------------|----------------------|--|--|---|--|---|---------------|
| 10015187 | E.I.S. AIRCRAFT GmbH | EASA.A.S.03571 | | | SOCATA TBM 700 | Avionics Upgrade to Garmin GTX330D diversity Mode S Transponder. | | Installation must be in accordance with EIS Aircraft GmbH Master Document List MDL 70S21010 Iss.00, dated 27/07/08 or later approved revisions.Operation must be in accordance with EIS Aircraft GmbH Flight Manual Supplement FM 70S21010-91501 Issue 00. (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.The I (...) | 14.08.2008 |
| 10015188 | E.I.S. AIRCRAFT GmbH | EASA.A.S.03572 | | | BEECH C90 A BEECH C90KING AIR | Installation of dual Rockwell/Collins TDR 94D diversity mode-s transponder. | | Installation must be in accordance with E.I.S. Aircraft GmbH Master Document List MDL 90C21011 Issue 02, dated 19.12.08or later Approved revisions.Operation must be in accordancewith E.I.S. Aircraft GmbH Flight Manual Supplement 90C21011-91501 Issue (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.The I (...) | 19.12.2008 |
| 10013951 | E.I.S.A. | EASA.A.S.00439 | REV 1 | | CESSNA 551 | Installation of two Rockwell / Collins TDR 94D instead of two TDR 90 according to FDM EISA 05FDM F055 rev. 4 dated November 2004 and associated documents.New transponder location | | | 1. Compatibility of this modification with other previouslyapproved modifications must be determined by the installer.2. Applicability Cessna 551 S/N 551-0133 and S/N CE551-0174.3. EASA approved Airplane Flight Manual Supplement, dated 23November 20 (...) | 30.10.2006 |
| 10013970 | E.I.S.A. | EASA.A.S.00741 | REV. 2 | | BEECH 200 | Photo Conversion / based on existing STC No. EASA.A.S.00741r1 | | EISA-16-MDL-F050-Rev1 Master Data List Issue 0 dated 17/12/08EISA-16-JSL-F050-Rev5 Structural Analysis Issue 5 dated 17/12/08PP-08-SA1-00 Analysis of optical window Issue 0 dated 11/12/08PP-08-LT1-00 Verification of Stress Level Issue 0 dated 14/12/ (...) | 1. The application of this STC is limited to the S/N BB221 and BB16072. Utilisation of Intercom prohibited in densetraffic areas3 Prior to installation of this modification the installer must determine that the interrelationship between this (...) | 27.02.2009 |
| 10014075 | E.I.S.A. | EASA.A.S.01167 | REV 1 | | FALCON 10 | Installation of a TCAS 4000 unit from Rockwell/CollinsInstallation of a TCAS 4000 unit from Rockwell/Collins | | - Data List EISA 06 SDL S085 Rev. 7- AFM supplement EISA 06AFM S085 Rev. 4- Antenna Installation according to the EISASTC no. EASA.A.S.0070- Data List EISA 06 SDL S085 Rev. 7- AFM supplement EISA 06 AFM S085 Rev. 4- Antenna Installation according to the EISA STC no. EASA.A.S.0070 | | 30.06.2006 |
| 10014530 | E.I.S.A. | EASA.A.S.02615 | | | CESSNA 404 | Installation of a video under Cessna 404 | | EISA AFM-S Doc. No S109, Revision 1, dated 13. June 2006, or later EASA approved RevisionEISA STC Type design definition EASA 00 STDD S109 Revision 1 or later EASA approved Revision | Applicable to Cessna 404 | 04.10.2006 |
| 10014643 | E.I.S.A. | EASA.A.S.02795 | | | MYSTERE FALCON 20 | Addition of Enhanced Surveillance Capacity on existing TDR 94 Mode S Transponder installation. | | Supplemental Type Certificate Data List EISA 05 SDL S114 Rev. 4Aircraft Flight Manual Supplement EISA 05 AFM S114 Rev. 1Aircraft Maintenance Manual Supplement EISA 05 AMMS S114 Rev. 0Or later approved revisions of these documents. | Major Design Change applicable to aircraft Fan Jet Falcon and Falcon 20 equipped with:# Rockwell Collins Proline IV avionics suite.# Mode S Elementary Transponder TDR-94D-008.# FMS-6000.# Air Data Computer ADC-850C.# An Input/Output C (...) | 01.02.2008 |
| 10014736 | E.I.S.A. | EASA.A.S.02919 | REV 1 | | CESSNA 525A | Installation of SAT-COM Antenna | | EISA 01 SDL S120, EISA 01 STDD S120 | | 24.01.2007 |
| 10014786 | E.I.S.A. | EASA.A.S.02986 | | | FALCON 50 | ADDITION OF ENHANCED SURVEILLANCE CAPACITY ON EXISTING ELEMENTARY MODE S TRANSPONDER INSTALLATION | | Compliance Checklist EISA 05 CCL S119 Revision 2 or later approved revisions.Supplemental Type Certificate Data List EISA 05 SDL S119 Revision 2 or later approved revisions.Aircraft Flight Manual Supplement EISA 05 AFM S119 Option 1 revision 1 or la (...) | This STC is approved only for those Mystere Falcon 50 aircraft previously equipped to this modification with:# Option 1: Rockwell Collins Proline IV, FMS FMC-6000 and Mode S Elementary Surveillance Installation with Transponder RockwellCollins TD (...) | 29.05.2007 |
| 10014908 | E.I.S.A. | EASA.A.S.03151 | | | FALCON 10 | MODE S TRANSPONDER WITH ENHANCED SURVEILLANCE CAPACITY | | | | 31.05.2007 |
| 10015021 | E.I.S.A. | EASA.A.S.03327 | | | CESSNA 525B | S134 - Maximum Takeoff/Landing Weight reduction. | | STC Data List (SDL) EISA 00 SDL Rev. 4Aircraft Flight Manual Supplement EISA 00 AFMS S134 Rev. 4 | None | 05.02.2008 |
| 10015364 | E.J. MLYNARCZYK & CO. INC. | EASA.IM.A.S.00679 | REV. 1 | FAA STC ST03078AT | BOEING 747-400 | Installation of a Panasonic Avionics Corporation System 3000 In-Flight Entertainment System in accordance with Master Drawing List MDL 10413, Rev. F dated 14 September 2005. Validation of FAA STC ST03078AT amended 22 September 2005. | | FAA-approved AFM supplement, document No. 14526015, dated 29 March 2005, or later FAA approved revision is a required part of this STC. | 1. STC No ST01523SE must be accomplished concurrently withMAS10413-01 and MAS10413-02 installations.2. This STC is approved only for the product configuration as defined in the approved designdata referred to in the paragraph "Description". Com (...) | 12.10.2005 |
| 10015856 | E.J. MLYNARCZYK & CO. INC. | EASA.IM.A.S.01632 | | ST01519AT | BOEING DC-9-82 (MD-82) BOEING DC-9-83 (MD-83) | FAA STC ST01519ATInstallation of Dual Canadian Marconi Company CMA-900 Flight Management System | | EJM Drawing List 40710010 Rev. G date 15th December 200AFMS40716029 Rev. B dated 21st February 2001Maintenance Advisory 40716031 dated 3 July 1997 | Compatibility of this design with previously approved modifications must be determined by the installer. | 31.07.2006 |
| 10039040 | E4C AEROSPACE, Inc. | | REV. 1 | FAA STC ST02110LA | B767-300 | Seat power, Seat Modifications and Seat track Installation | The Certification Basis for the original product remains applicable tothis certificate.The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this approval. | E4C master Data List 1063-000, revision D, 21/11/2007or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequentrevisions ofthis decision) | None | 15.05.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------|----------------|----------------|----------------------|--|---|---|---|--|---------------|
| 10013929 | EAD AEROSPACE | EASA.A.S.00265 | REV. 1 | | BOEING 737-300, 737-400, BOEING 737-500 | Installation of Honeywell EGPWS Mark V | | EAD Master Drawing List ref. 00772-JAA-A-MDL revision GEAD AFM Supplement ref. DC113-0772-00 revision HEAD MPD Supplement ref. DC114-0772-00 revision Aor later EASA approved revisions of those documents | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 02.02.2009 |
| 10014201 | EAD AEROSPACE | EASA.A.S.01351 | REV 1 | | SA 227 SERIES | ELS/EHS installation on SA 227 series aircraft- Removal of existing two ATC transponders and associated trays in forward avionics- Removal of existing ATC Mode S Control Panel from Instrument Panel- Disconnection of ATC 2 Mode S bottom antenna- Inst (...) | | Aircraft Flight Manual Supplement:Document 01046-EAS-A-AFM-01 Rev. BMinimum Equipment List Supplement:Document 01046-EAS-A-MEL-01 Rev. AMaintenance Planning Document Supplement:Document 01046-EAS-A-MPD-01 Rev: B | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 27.06.2006 |
| 10014450 | EAD AEROSPACE | EASA.A.S.02503 | | | B757 | ELT antenna installation as provision only | | Compliance Check List (reference 01060-EAS-C-CCL-01, Revision A)Certification Program (reference 01060-EAS-C-CPM-01, Revision A)Master Drawing List (reference 01060-EAS-C-MDL, Revision A)Maintenance Planning Document Supplement (reference 01060-EAS- (...) | Maintenance Planning Document Supplement 01060-EAS-C-MPD-02, Revision A | 10.03.2006 |
| 10014637 | EAD AEROSPACE | EASA.A.S.02783 | REV 1 | | BOEING 767 | Installation of Panasonic "EFX" Passenger in Flight Entertainment System (IFE). | | EAD Aerospace Master Drawing List 01110-EAS-A-MDL at Revision G.Airplane Flight Manual Supplement 01110-EAS-A-AFM-01 atRevision C. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 04.04.2007 |
| 10014638 | EAD AEROSPACE | EASA.A.S.02784 | | | BOEING 737-600 BOEING 737-700 BOEING 737-800 BOEING 737-900 | Implementation of Honeywell #Link 2000# Controller-Pilot Data Link Communications (CPDLC) on B 737 | | - Supplemental Type Design Definition Doc 01106-EAS-A-STD-01 Revision C, or later EASA approved Revision - Master Drawing List 01106-EAS-A-MDL Revision E or later EASA approved Revision.- AFM Supplement 01106-EAS-A-AFM-01 RevisionB, or later (...) | Compatibility of this design change with other previously approved modifications must be determined by the installer. | 09.03.2007 |
| 10014664 | EAD AEROSPACE | EASA.A.S.02821 | | | BOEING 757-200 SERIES | Installation of Thales Integrated Standby Flight System | | Certification Plan 01109-EAS-A-CPM-01 Rev BMaster Drawing List 01109-EAS-A-MDL Rev CGround Test Plan/Report 01109-EAS-A-GTR-01 Rev BCompliance Checklist 01109-EAS-A-CCL-01 Rev AEAD AFM Supplement 01109-EAS-A-AFM-01 Rev BEAD MEL Supplement 01109-EAS-A-MEL-01 Rev B | | 11.05.2007 |
| 10014679 | EAD AEROSPACE | EASA.A.S.02840 | | | B767-300 | Installation of Thales Integrated Standby Flight Display (ISFD) | | EAD Aerospace 01108-A,Flight Manual Supplement 01108-EAS-A-AFM-01 Rev A or later EASA Approved Revision. | | 05.03.2007 |
| 10015106 | EAD AEROSPACE | EASA.A.S.03449 | | | A319 | Installation of AMT-3800 High Gain Antenna on A319 Series Aircraft | | 2. Master Drawing List, ref. Doc. No. 01227-EAS-A-MDL, Rev. D, dated 14th November 2008; 3. Supplemental Type Design Document, ref. Doc. No. 01227-EAS-A-STD-01, Rev. B, dated 5th August 2008; 4. Maintenance Planning Document, ref. Doc. No. 01227-E (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 17.11.2008 |
| 10039303 | EAD AEROSPACE | | REV. 1 | | B747-400 | Installation of Cobham SATCOM antenna type HGA-7001 in accordance withAerotec Master Data List AE11110MDL Rev.1, dated 4 April 2012. | The certification basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product areunchanged and remain applicable to this certificate/approval. | AE11110MDL Rev. 1 Aerotec Master Data List, dated 4 April 2012AE11110SB Instructions for Continued Airworthiness, dated14 December2011, or later revisions of the above listed documents approved by EASA. | None. | 03.07.2012 |
| 10040043 | EAD AEROSPACE | | REV. 1 | | B747-400 | Swift broadband Satcom installation in accordance with Aerotec ConceptMaster Data List AE11111MDL Rev 2, dated April 3,2012 | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approval.CS-25 Arndt. 1 for areas of changeThe requirements for environmental protection and the (...) | Master Data List AE11111MDL Rev 2, dated April 3, 2012or later revisions of the above listed documents approved by EASA | S/N: 30865, 32337, 32338, 32339, 32340, 28757, 29406 | 06.07.2012 |
| 10043418 | EAD AEROSPACE | | REV. 1 | | A340-541 | Description for the initial approval:Installation of GSM System in an A340GSMConnex is solution that provides GSM services to aircraft passengersand crew. It consists of hardware and software parts for both theaircraft and the ground, withthe airbor (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Documentation related to the initial approval not totally superseded bythe revision A:Compliance Check List 01388-EAS-A-CCL-01, Issue D, dated 24th January2013.Documentation related to revision A:Airplane Flight Manual 01388-EAS-A-AFM-01, Rev.C, date (...) | This approval is limited to the aircraft MSN 495 (A340-541)configuredaccording to the ICA document 01388-EAS-A-ICA-01-C. As is stated in thisdocument the aircraft must be equipped only with those elements provided to be compatible with the GSMConnex in (...) | 06.05.2013 |
| 10045202 | EAD AEROSPACE | | | | A330-202 | GSMConnex is solution that provides GSM services to aircraft passengersand crew. It consists of hardware and software parts for both theaircraft and the ground, withthe airbor (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associatedcertified noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Airplane Flight Manual Supplement 01383-EAS-A-AFM-01, IssueC, dated29th May 2013.Master Drawing List 01383-EAS-A-MDL, Issue D, dated 29th May 2013.Instructions for Continued Airworthiness, 01383-EAS-A-ICA-01 Issue C,dated 30th May 2013.Compliance Ch (...) | This approval is limited to the aircraft MSN 605 (A330-202)configuredaccording to the ICA document 01383-EAS-A-ICA-01 Issue C. As is stated in this document the aircraft must be equipped only with those elements provided to be compatible withthe GSMConn (...) | 06.06.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|-------------------|----------------|----------------------|---|---|--|---|---|---------------|
| 10045334 | EAD AEROSPACE | | | | B737-600/-700/-800/-900/900ER | Upgrade to CDPLC for ATN B1 capability upgrade of existing Communication Management Unit software for CPDLC (Pioneer version of the Rockwell-Collins solution) with an ATN B1 compliant version. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:- Special Condition(s): F-01 Data Link. Services for the Single European Sky Issue 02 dated 26.03.2013 - the following paragr (...) | Master Drawing List 01406-EAS-A-MDL-B Aircraft Flight Manual Supplement 01406-EAS-A-AFM-01-A or later revisions of the above listed documents approved by EASA. | As per Aircraft Flight Manual Supplement 01406-EAS-A-AFM-01-A: "Except for following notified hereafter, the CPDLC installation is limited to providing a supplementary means of communication; voice shall remain the primary means of communication. CPDLC sh (...) | 17.06.2013 |
| 10045456 | EAD AEROSPACE | | | | 777-300/300ER | Activation of Cabin Connectivity System (CCS). [GSM/ConneX# end-to-end solution providing GSM and WIFI services to aircraft passengers.] | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Drawing List, EAD Aerospace document No. 01410-EAS-A-MDL, Revision D (>-<)-> dated 25 June 2013. - Aircraft Flight Manual Supplement, EAD Aerospace document No. 01410-EAS-A-AFM-01, Revision C, dated 24 June 2013 - Instruction for Continued Airworth (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the produc (...) | 26.06.2013 |
| 10045906 | EAD AEROSPACE | | | | 737-600, 737-700, 737-800, 737-900, 737-900ER. | Activation of the GSM capability using the Triagnoeys GSM Connex solution. | The applicable certification basis (CB) is the one of the B737-900ER (JAR 25 change 15). The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List 01397-EAS-A-MDL rev C Aircraft Flight Manual Supplement 01397-EAS-A-AFM-01 rev A Instructions for Continuous Airworthiness Supplement 01397-EAS-A-ICA-01 rev Cor later revisions of the above listed documents approved by EASA. | - This approval is limited to the aircraft configurations identified in document 01397-EAS-A-AEL-01 (last revision)- Airworthiness limitations as per 01397-EAS-A-ICA-01 rev CPrior to installation of this design change it must be determined that the in (...) | 26.07.2013 |
| 10015302 | EAD AEROSPACE AIRWORTHINESS | EASA.A.S.03866 | REV. 3 | | A340 | Installation of Advanced Mission Display System Phase A- AMDS/Revision of the EASA STC EASA.A.S.03866 Revision 2 to remove EWIS limitation | Certification Basis in accordance with EASA Type Certificate Data Sheets A.015. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. 1. CRI H (...) | - Master Drawing List, ref. 01274-EAS-A-MDL, Revision K, dated 26th November 2010- Compliance Check List, ref. 01274-EAS-A-CCL-01, Revision B, dated 18th June 2009- Airplane Flight Manual Supplement, ref. 01274-EAS-A-AFM-01, Revision D, dated 18th June (...) | Not applicable | 02.12.2010 |
| 10027343 | EAD AEROSPACE AIRWORTHINESS | P-EASA.A.S.03706 | | | A330 | Installation of Advanced Mission Display System - Phase A (AMDS/A) | In accordance with EASA TCDS A.004 | - Master Drawing List, ref. 01250-EAS-A-MDL, Revision C, dated 22nd September 2009- Design Similarity Statement, ref. 01250-EAS-A-DSS-01, Revision A, dated 31st July 2009- Airplane Flight Manual Supplement, ref. 01250-EAS-A-AFM-01, Revision A, dated 31 (...) | None | 23.09.2009 |
| 10031354 | EAD AEROSPACE AIRWORTHINESS | | REV. 1 | | A320 | Installation of Advanced Mission Display System Phase A- AMDS/A on A320 Series Aircraft/Revision of the EASA STC 10031354 to remove EWIS limitation | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- CRI H-01/Special Condition H-01 "Enhanced Airworthines (...) | - Master Drawing List, ref. 01272-EAS-A-MDL, Revision C, dated 22nd November 2010- Compliance Check List, ref. 01272-EAS-A-CCL-01, Revision A, dated 9th April 2010- Airplane Flight Manual Supplement, ref. 01272-EAS-A-AFM-01, Revision A, dated 9th April (...) | None | 29.11.2010 |
| 10036787 | EAD AEROSPACE AIRWORTHINESS | | | | A320 SERIES | Cabin Modification 174 to 168 PAX on A320 | In accordance with the current EASA TCDS The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of th (...) | - Master Drawing List, ref. 01346-EAS-A-MDL, Revision C, dated 4th October 2011- Compliance Check List, ref. 01346-EAS-A-CCL-01, Revision C, dated 4th October 2011- Supplemental Type Design, ref. 01346-EAS-A-STD-01, Revision B, dated 4th October 2011- (...) | None | 06.10.2011 |
| 10043327 | EAD AEROSPACE AIRWORTHINESS | EASA.A.S.03881 | | | AIRBUS A320 | Installation of AMT-3800 High Gain Antenna on A320 | | - Master Drawing List, ref. Doc. No. 01273-EAS-A-MDL, Rev. D, dated 23rd July 2009- Supplemental Type Design Document, ref. Doc. No. 01273-EAS-A-STD-01, Rev. B, dated 15th May 2009- Maintenance Planning Document, ref. Doc. No. 01273-EAS-A-MPD-01, Rev. C, d (...) | Limitations and Conditions: 1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthi (...) | 30.07.2009 |
| 10045413 | EAD AEROSPACE AIRWORTHINESS | | | | A320-111/-211/-212/-214/-215 A320-216/-231/-232/-233 | GSM Activation | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification Program 01381-EAS-CPM-01 Rev C, dated 01/03/2013 Compliance Check List 01381-EAS-A-CCL-01 Rev A, dated 12/06/2013 Master Drawing List 01381-EAS-A-MDL Rev C, dated 12/06/2013 Aircraft Flight Manual Supplement 01381-EAS-A-AFM-01 Rev A, date (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the produc (...) | 21.06.2013 |
| 10016076 | EAD INC. | EASA.IM.A.S.02260 | | FAA STC ST02267CH | BOEING 747-400 | EAD installed Honeywell Weight and Balance System (WBS) on B747-400 in accordance with EAD Master Data List 01021-FAA-A-MDL, rev. E, or later EASA-approved revision. This STC is a validation of FAA STC ST02267CH | | Installation of Weight and Balance System on B747-400, Master Data List, EAD document 01021-FAA-A-MDL, Rev. E, Airplane Flight Manual Supplement, EAD document 01021-FAA-A-AFM-01, Rev. D. Installation of Weight and Balance System on B747-400 | 1. This on board weight and balance system is considered an advisory system and is not certified as the primary means of calculating the aircraft's weight and balance. 2. EASA approved Airplane Flight Manual Supplement, EAD document 01021-FAA-A-AFM-0 (...) | 27.04.2007 |
| 10016545 | EAD INC. | EASA.IM.A.S.02924 | | FAA STC ST01481CH | BOEING 737-400 SERIES | TCAS/Mode-S Activation on Boeing 737-400 Series as approved under the STC FAA ST01481CH. | | 1. Europe Aero Design Master Data List, Doc. No. DC104-0245-00 Rev.2, dated June 26, 2001** - FAA approved Airplane Flight Manual Supplement, Europe Aero Design Doc. No. DC353-1493-00 Rev.0 approved June 27, 2001** or later revisions of the above (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 02.10.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|------------------|----------------|----------------------|---|---|---|---|---|---------------|
| 10034640 | EAD INC. | EASA.M.A.S.01715 | REV. 1 | FAA STC ST03205AT | BOEING DC-9-82 (MD-82) | Installation of Rockwell Collins CMU-900 Communications Management Unit,VHF-920 VDL Mode 2 Radio, Gables Mode 2 VDR Control Panel and ACARS TP4429 Printer to provide Controller Pilot Data Link Communications(CPCDL) Baseline 1 service in accordance w (...) | | -EAD Aerospace Aircraft Flight Manual Supplement, Document No.01049-EAS-B-AFM-01 rev. B-FAA STC ST03205AT, reissued 27 January 2009. | None | 31.07.2009 |
| 10045910 | EAD INC. | | | FAA STC ST02134CH | MD-11 MD-11F | ELT Installation.Installation of an ELTA ADT 406 AF/AP or ELTA A06V2 Emergency Locator Transmitter. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | For Emergency Locator Transmitter ELTA ADT 406 AF/AP in Model MD-11F:- Certification Plan DC103-0670-00, Revision A, 26 August 2004- Master Data List 00670-FAA-A-MDL, Revision C, 21 November 2005<->->- AFM S DC304-0670-00, Issue A, 28 January 2005<- (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 26.07.2013 |
| 10029272 | EADS CASA | P-EASA.A.S.02790 | REV. 3 | | A330-203 WEIGHT VARIANT 052 FOR CONFIGURATION 1 A330-202 WEIGHT VARIANT 052 FOR CONFIGURATION 2 A330-243 (WEIGHT VARIANT 052) FOR CONFIGURATION 3 | Conversion of A330-200 Series A/C to A330 Multi Role Transport Tanker(MRTT) RAAF (Royal Australian Air Force), RSAF (Royal Saudi Arabian Air Force) and UAE (United Arab Emirates Air Force) Rev. 1 of this STC has been issued in order to add Configuration (...) | The Certification Basis for the original product remains applicable to this certificate/ approval.This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certificated noise and/or emissions levels | For Configuration 1 RAAF:- "Aircraft A330 MRTT Type Design Definition for STC" document ref.DT-FA-ANC-09077 Issue A- Airplane Flight Manual Supplement "EASA Approved A330 MRTT Airplane Flight Manual" document ref. DT-FA-C00-08025 For Configuration 2 R (...) | Approval of this STC is limited to compliance to selected Certification Standards as defined in CRI A-01. CRI G-01 defines in particular restrictions relating to Maintenance Tasks and Airworthiness Limitations, ref JAR 25.571a(3) and 25.1529. Therefore, (...) | 21.12.2011 |
| 10034690 | EADS CASA | P-EASA.A.S.03608 | REV. 2 | | A330-243 | A330 MRTT FSTA Modification STC1 With the Revision 1 of this STC a previous limitation on the occupational medical stretchers has been deleted. With the revision 2 of this STC the role change capability is introduced. Therefore, it is possible for an (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | "Conversion Definition A330 MRTT FSTA" document ref. ME-FU-DT-110145 Issue A- ALS Repository "A330 FSTA EASA STC1 Airworthiness Limitation Section" document ref. DT-FU-C00-10019 Issue C- ALS Part 1 Supplement "A330 FSTA Safe-Life Airworthiness Limita (...) | None | 02.08.2012 |
| 10035945 | EADS CASA | P-EASA.A.S.03607 | REV. 3 | | A330-243 | A330 FSTA Extension of the ETOPS capability of the Basic Aircraft to STC10035945 Revision 2 A330 MRTT FSTA Modification STC2 Revision 1 Remark: Revision 1 of the STC introduces the following changes:- Deletion of previous limitation no. 4 on Autoland / (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | "Airbus Military Document Ref. ME-FU-CT 130055 dated 04.09.2013- A330 FSTA Airplane Flight manual TR 4- "Conversion Definition A330 MRTT FSTA (STC2)" document ref. ME-FU-DT-110148 Issue D- ALS Part 1 Supplement "A330 FSTA Safe-Life Airworthiness Limi (...) | Based on the conditions established in CRI G-01 with respect to the definition of the Airworthiness Limitation Section of the Instructions for Continued Airworthiness the following limitations apply:1. As the modifications of this STC are intended for (...) | 05.09.2012 |
| 10042436 | EADS CASA | | | | A330-243 | Installation of a Cockpit Door Surveillance System in the A330 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 1. Airbus Military Document Ref. ME-FU-DT-120055, dated 29/10/2012 "A330 Cockpit Door Surveillance System Certification" 2. EADS - CASA Declaration of Compliance for the EASA Project 0010019789-001 dated 29.11.2012 | None | 30.11.2012 |
| 10014980 | EADS ELBE FLUGZEUGWERKE GmbH | EASA.A.S.03272 | | | AIRBUS A310-304 | A310-304 Cabin Reconfiguration/Transfer of the STC holder, from ASL Aircraft Services Lemwerder GmbH to EADS Elbe Flugzeugwerke GmbH. Former LBA STC ref TA0721 Cabin Reconfiguration from 265 seats to 225 seats (18 Business Class + 207 Economy Class). | | Certification Report, ref : ASL CR 310-25-0001, dated 11 April 2003 | 1. This STC is limited to A310-304 aircraft model MSN 448.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce n (...) | 08.10.2007 |
| 10014981 | EADS ELBE FLUGZEUGWERKE GmbH | EASA.A.S.03273 | | | AIRBUS A310-304 | A310-304 Floor Path Marking System installation/Transfer of the STC holder, from ASL Aircraft Services Lemwerder GmbH to EADS Elbe Flugzeugwerke GmbH. Former LBA STC ref TA0719 The modification related to this supplement concerns the installation of a photoluminescent emergency escape path marking. | | Certification Report, ref : ASL CR 310-33-0001, dated 10 April 2003 Aircraft Flight Manual Supplement, ref : AFMS310330001, dated 06 March 2003 | 1. This STC is limited to A310-304 aircraft model MSN 448.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce n (...) | 08.10.2007 |
| 10014982 | EADS ELBE FLUGZEUGWERKE GmbH | EASA.A.S.03274 | | | AIRBUS A320-214 | A320-214 Cabin Reconfiguration/Transfer of the STC holder, from ASL Aircraft Services Lemwerder GmbH to EADS Elbe Flugzeugwerke GmbH. Former LBA STC ref TA0717 Issue 2 Cabin Reconfiguration from 156 seats (90 Business Class + 66 Economic Class) to 180 seats (Economy Class). | | Certification Report, ref: ASL CR 320-25-0001 Revision B, dated 20 May 2003 | 1. This STC is limited to A320-214 aircraft model MSN 1081 and MSN 1370.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification wi (...) | 08.10.2007 |
| 10031745 | EADS ELBE FLUGZEUGWERKE GmbH | P-EASA.A.S.03661 | | | A300B4-601/ -603/ -620/ -622 A300B4-605R/ -622R A310-304/ -308/ -324/ -325 | Courier Stowage installation This STC consists in modifying the Courier Area of aircraft converted by Airbus Pax to Freighter STCs by installing a Courier Stowage | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS25 Amendment 4 Complementary Condition GC12-1; Crash design conditions Special Conditio (...) | Engineering Order, ref: EFW-EO-0020-001, Issue B or later revisions of the above listed documents approved by EASA | The current STC needs specific aircraft configuration as prerequisite: - To install this STC on A310 aircraft models, embodiment of EASA STC ref: 10013945, Rev 4 (or any further EASA approved revision) is required. - To install this STC on A300-600 air (...) | 09.09.2010 |
| 10013940 | EADS SOGERMA | EASA.A.S.00376 | REV 1 | | AIRBUS A310-304 | Royal Jordanian ATC Mode S Enhanced Surveillance installation according to EADS SoGerma Modification Approval Sheet SOG-34-974 issue 3 dated 29 November 2004. Note: This revision reflects the transfer of this STC from EADS SoGerma Services to SoGerma Services. | | | 1. The Aircraft Flight Manual Supplement No AFM-34-974 issue B dated 29 November 2004, or later EASA-approved revision, is a required part of this STC. 2. Prior to installation of this modification the installer must determine that the interrel (...) | 07.01.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------|----------------|----------------|----------------------|------------------------------------|---|---------------------|---|--|---------------|
| 10013941 | EADS SOGERMA | EASA.A.S.00377 | REV 1 | | AIRBUS 310-325 | Cabin Modification from 18F/C-180Y/C to 18B/C-188Y/C according to the modification approval sheet MAS No SOG-25-957 issue 1 dated 6 October 2004 or later EASA-approved revision.Note: This revision reflects the transfer of this STC from EADSSogerma Services to Sogerma Services. | | | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 07.01.2008 |
| 10013950 | EADS SOGERMA | EASA.A.S.00432 | REV 1 | | AIRBUS 310-325 | Modification of ATLAS/KSSU galleys into full ATLAS standard in accordance with Modification Approval Sheet reference SOG-25-958 Issue 01 dated 14 October 2004. All documents related to this STC are listed in the Master Drawing List reference MDL-25-9 (...) | | | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 07.01.2008 |
| 10014005 | EADS SOGERMA | EASA.A.S.01042 | REV 1 | | AIRBUS A340-300 | Installation of an upper class with side facing Britax Malcom seats with inflatable restraints, monument, bar and chandelier, and modification of the premium and economy class by repitching.Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services. | | MAS SOG-00-1000 Issue 1 dated 1/08/2005.Master Data List Drawing No. MDL 00-1000 Issue No. A, or later EASA approved revision.Flight Manual Supplement page 1.02.00 p01 dated 01/07/2005 .Technical repercussion sheet Ref. 00-1000A. | 1. This STC is limited to Airbus A340-300 MSN 214, 002, 003, 164 and 225.2. A copy of this STC and of the MAS/MDL must be enclosed in the files of each modified aircraft. | 07.01.2008 |
| 10014022 | EADS SOGERMA | EASA.A.S.01077 | REV 1 | | AIRBUS A330-203 | Qatar Airways - Cabin Modification. Modification number: SOG-25-886 issue 1.Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services.Model: Airbus A330-203 MSN 463 and 504 | | -AFM Supplement ref.: AFM-25-886 Iss. A | 1. The approval of this modification is limited to the identified aircraft. The limitations and conditions defined in the EASA Datasheet are also applicable. A copy of the STC approval and the MDL must be enclosed in the files of each modified aircraft (...) | 07.01.2008 |
| 10014076 | EADS SOGERMA | EASA.A.S.01168 | REV 1 | | AIRBUS A330-243 | Cabin configuration for Gulf Air. All documents related to this STC are listed in the Modification SOG-25-1075 Iss. 01 (or later EASA-approved revision).Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Servi (...) | | Modification SOG-25-1075 Iss. 01 (or later EASA-approved revision) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 07.01.2008 |
| 10014077 | EADS SOGERMA | EASA.A.S.01169 | REV 1 | | AIRBUS A310 | Installation of an ATC Mode S EH5. ATC mode S modification for compliance with Enhance Surveillance. Replacement of the Rockwell Collins TRP-720 ATC by TRP-901 ATC.Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services. | | Modification Approval Sheet SOG-34-866 Issue 02. Aircraft Flight Manual Supplement: AFM-34-52-00 Issue A dated 15/07/2004. Compliance Statement CS-34-866-01 Issue A. | | 07.01.2008 |
| 10014087 | EADS SOGERMA | EASA.A.S.01179 | REV 1 | | DC-9-21 | ELT Kannad 406 AF modification (SOG-25-1106).Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services. | | Airplane Flight Manual Supplement ref FMS 25-1106 issue C, dated November 11, 2005 | 1. This STC is limited to the DC-9 MSN 4708 | 04.01.2008 |
| 10014113 | EADS SOGERMA | EASA.A.S.01219 | REV 1 | | ATR 42-320 SERIES | Pseudo-Lisse configuration; RESSAC configuration and RALI configuration.Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services.Model: Aerospatiale ATR 42-320 MSN 078 | | AFM Supplement for Pseudo-Lisse configuration referred 238/04/SOG/DEV Issue 7/AFM Supplement for RESSAC configuration referred 239/04/SOG/DEV Issue 2.AFM Supplement for RALI configuration referred 240/04/SOG/DEV Issue 2.Weight & Balance Supplement for (...) | 1. In RALI configuration: the mass of under wing POD is limited to 72kg (POD equipped); the new design for the titanium pin on POD rear attachment must be applied according to the EADS Sogerma Services waiver No. A3002.2. The lighting issues a (...) | 04.01.2008 |
| 10014114 | EADS SOGERMA | EASA.A.S.01220 | REV 1 | | AEROSPATIALE ATR 42-320 | ELT Kannad 406 AF Modification.Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services. Model: Aerospatiale ATR 42-320 MSN 078 | | Technical Repercussion Sheet referred SOG-25-1104 Master Drawing List MDI-25-1104, Issue No A dated February 2, 2005 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 04.01.2008 |
| 10014169 | EADS SOGERMA | EASA.A.S.01302 | REV 1 | | AIRBUS A321-200 | A321 Livingston - Cabin Modification from 196 Y/C to 210 Y/C with relocation of two attendant seats.Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services.Model: Airbus A321-200 MSN 1950 & 1970 | | Modification Approval Sheet (MAS) Ref. SOG-25-1102 Issue 01, dated May 16, 2005 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 04.01.2008 |
| 10014171 | EADS SOGERMA | EASA.A.S.01304 | REV 1 | | AIRBUS A330-200 | Installation of a Cabin modification from 24B/C - 248Y/C to 18B/C - 271Y/C with removal of galley G2 and installation of new partitions for LAUDA AIR fleet. The associated modification Approval sheet is referenced SOG-25-1122 issue1 dated 10/06/05. A (...) | | Technical repercussion sheet reference 25-1122. | | 04.01.2008 |
| 10014675 | EADS SOGERMA | EASA.A.S.02835 | REV 1 | | AIRBUS A320-211 AIRBUS A320-212 | Emergency locator transmitter Thales/Serpe Kannad 406 ATP installation.Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services. | | Airplane Flight Manual Supplement ref. FMS 25-967, issue B, dated November 3, 2004 or later EASA approved revision. | | 04.01.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|----------------------|--|--|---------------------|---|--|---------------|
| 10043109 | EADS SOGERMA | EASA.A.S.02569 | | | DC-9-21 | JANUS Phase II modification | | Master drawing list, ref. MDL 01-S67, issue B, dated March 4, 2005 Airplane Flight Manual Supplement, ref. DIITNRJ4084/05, issue 2, dated April 7, 2005 | Eddy current inspections ref. MMSOS-537210 and MM808-53711 Oas defined in the Maintenance Schedule Supplement ref. MS 01-867 issue A, dated February 10, 2005. This STC is limited to the DC-9 MSN 47308. This STC is approved only for the product configuratio (...) | 14.12.2006 |
| 10016905 | EAGLE COPTERS MAINTENANCE Ltd. | EASA.IM.R.S.01284 | | TCCA STC SH92-28 | BELL 204B BELL 205A BELL 205A-1 | Dual Cargo Hook Mechanical release Kit. | | A20468DCL1 - Document Control List, Revision 0 dated 28 August 1991 or later approved revision; EA204-68-18 - Service Instruction, Revision 0 dated 29 August 1991 or later approved revision. | | 09.10.2007 |
| 10016906 | EAGLE COPTERS MAINTENANCE Ltd. | EASA.IM.R.S.01285 | | TCCA STC SH92-30 | BELL 204B | Vertical Reference Door. | | A20468DCL1 - Document Control List, Revision 9 dated 18 June 2007 or later approved revision; EA204-69-19 - Service Instruction, Revision 4 dated 28 April 1998 or later approved revision; EA204-69-20 - Flight Manual Supplement, Revision 1 dated 3 September 1992 or later approved revision. | Day VFR Only; Minimum Flight Crew - 1 in left crew seat; Crew restricted to - Flight Crew-Flight Crew under Training- Persons essential for the type of operation Dual controls must be installed per Bell Helicopter Service Instruction 206-6 for Model 20 (...) | 09.10.2007 |
| 10016977 | EAGLE COPTERS MAINTENANCE Ltd. | EASA.IM.R.S.01379 | | TCCA STC SH00-37 | BELL 204B | DCL 361 Tailboom Vertical Fin Spar Reinforcement. New STC for modification of tailboom vertical fin spar to comply with requirements of FAA AD 99-17-03. | | DCL361 - Aero Design Ltd. Document Control List - revision 5 dated 10 October 2000 or later approved revision; MMS361.03- Eagle Copters Ltd. Maintenance Manual Supplement - revision 2 dated 10 October 2000 or later approved revision. | This approval supports an Alternative Means of Compliance (AMOC) for satisfying FAA AD99-17-03. An EASA AMOC authority is required in addition to this STC. | 15.11.2007 |
| 10016980 | EAGLE COPTERS MAINTENANCE Ltd. | EASA.IM.R.S.01383 | REV. 1 | TCCA STC SH03-14 | BELL 204B BELL 205A-1 BELL 205B BELL 212 BELL 214B BELL 214B-1 BELL 412 BELL 412 CF BELL 412 EP | D412-758 GPS receiver Gimbal Mount | | 1) Master Document List MDL-D412-758 Revision D dated 8 April 2008(2) Installation Instructions IIN-D412-758 Revision B dated 8 April 2008(3) Instructions for Continued Airworthiness ICA-D412-758 Revision 2 dated 7 April 2008 | 1) Eligible GPS receivers are listed in Installation Instructions IIN-D412-758 Revision B dated 8 April 2008(2) GPS operation is limited to VFR use only | 21.04.2008 |
| 10017024 | EAGLE COPTERS MAINTENANCE Ltd. | EASA.IM.R.S.01442 | | FAA STC SR00600NY | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 C EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 EUROCOPTER FRANCE AS 355N | D350-721 Lightweight Heli Utility Basket Installations. | | IIN-D350-721 revision A dated 18 November 2004 or later approved revision # Installation Instructions.FMS-D350-721 revision B dated 3 June 2008 or later approved revision # Flight Manual Supplement; ICA-D350-721 dated 22 November 2004 or later approve (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 05.06.2008 |
| 10016904 | EATON AEROSPACE | EASA.IM.R.S.01281 | | | S-76A, S-76C | Installation of Eaton Aerospace/Tedeco Products Zapper pulsed electric chip detector system 1K7707 on Sikorsky Helicopter S-76A and C Models with Turbomeca Arriel 1S, 1S1, 2S1 and 2S2 Engines in accordance with Installation Drawing List IDL00508, Rev (...) | | see above | 1. For the Sikorsky S-76 helicopters fitted with Arriel 1S engines - the Arriel 1S and 1S1 engines must be equipped with TU 208 and TU 232 prior to the installation of the 1K7707 Zapper System.2. Eaton Aerospace Rotorcraft Flight Manual Supplement, (...) | 07.11.2006 |
| 10017099 | ECMS AVIATION SYSTEMS GMBH | EASA.R.S.01023 | | | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 BB EUROCOPTER FRANCE AS 350 D | Installation of External Load Box ELB01 | | -Installation Instructions ELB01_inst, issue 04.10.2005 or later approved revision-Flight Manual Supplement ELB01, issue 16.09.2005 or later EASA approved revision-Maintenance Manual ELB01_mm, issue 24.01.2006 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 30.01.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------|----------------|----------------|----------------------|--|---|---|---|--|---------------|
| 10017173 | ECMS AVIATION SYSTEMS GMBH | EASA.R.S.01288 | | | EUROCOPTER DEUTSCHLAND EC-135 | Balance Weighing System BWS02. | | Flight Manual Supplement BWS02_fms, issue 01-07-2007 or later approved revision. Installation instructions BW S02_inst.Maintenance Manual & User Guide BWS02_mm Ug. | The Balance Weighing System BWS02 is compatible with the following EC135 cargo hook installations: Dual External Cargo Hook (Eurocopter FMS 9.2-47) or External Cargo Hook (Eurocopter FMS 9.2-19). Fixed provisions including internal wiring harness are req (...) | 20.07.2007 |
| 10017200 | ECMS AVIATION SYSTEMS GMBH | EASA.R.S.01349 | REV. 2 | | EUROCOPTER DEUTSCHLAND EC-135- EUROCOPTER DEUTSCHLAND EC-635 EUROCOPTER DEUTSCHLAND EC-635- | "Load Attachment Device EAD01" Provides attachment provisions on rotorcraft for fast roping / rappelling and Human External Cargo operations. Revision 1 to the STC is a revision of the previous LBA STC No. 1025, dated 24.11.2003. Revision 1 adds appli (...) | | # Flight Manual Supplement for EAD01, Revision 4, dated 01.10.2008# Maintenance Manual/User Guide EAD01_MM, Revision 4, dated 01.10.2008 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 11.12.2008 |
| 10017201 | ECMS AVIATION SYSTEMS GMBH | EASA.R.S.01350 | REV. 1 | EASA STC 1025 | EUROCOPTER DEUTSCHLAND BK117 C | #External Absell Device EAD01#Provides attachment provisions on rotorcraft for fast roping / rappelling and Human External Cargo operations. This STC is a revision of the previous EASA STC No. 1025, dated 24.11.2003. The Technical Changes areas follows (...) | | # Flight Manual Supplement for Fixed External Absell Device on EC145 (BK 117 C-2), Revision 2, dated 01.01.2007.#Maintenance Manual/User Guide EAD01_MM, Revision 1 dated 01.01.2007. | 1. The scope of this STC is limited to attachment provisions for rappelling / fast roping and HEC operations. It does not include all the necessary equipment like ropes and personal carrying device. It does not establish all the procedures necess (...) | 30.01.2007 |
| 10017250 | ECMS AVIATION SYSTEMS GMBH | EASA.R.S.01451 | | | EUROCOPTER FRANCE AS 365N3 EUROCOPTER FRANCE EC 155 B1 EUROCOPTER FRANCE EC 155B EUROCOPTER FRANCE SA 365C2 | External Load Attachment Device EDX02. | | Flight Manual Supplement EDX02_FMS- EASA, Original Issue, 01.05.2005. Installation Instructions EDX02_INST_EASA/Maintenance Manual & user Guide EDX02_MM_UG_EASA. | The scope of this STC is limited to technical provisions to the helicopter. It does not include ropes and personal equipment, training aspects and operation procedures necessary for the fast roping operation. These procedures must be established by t (...) | 18.07.2008 |
| 10017273 | ECMS AVIATION SYSTEMS GMBH | EASA.R.S.01503 | | | EUROCOPTER DEUTSCHLAND EC-135- EUROCOPTER DEUTSCHLAND EC-635 EUROCOPTER DEUTSCHLAND EC-635- | Installation of HEC Rope System HRS01, P/N HRS01-01-00.00 | | Flight Manual Supplement, Doc No HRS01_fms_EC135_EASA, dated 25.06.2008 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 11.12.2008 |
| 10017282 | ECMS AVIATION SYSTEMS GMBH | EASA.R.S.01530 | | | EUROCOPTER DEUTSCHLAND MBB-BK | Installation of HEC Rope System HRS01, P/N HRS01-01-00.00 | | Flight Manual Supplement, Doc No. HRS01_FMS_BK117_EASA_R0, dated 25.06.2008 | | 19.12.2008 |
| 10017283 | ECMS AVIATION SYSTEMS GMBH | EASA.R.S.01531 | | | EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 BB EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350B | External Antenna for Transceiver System ATSO1 installation on the Idg gear crosstube. | | FM Supplement for Aerial for Transceiver System ATSO1 # doc. No. ATSO1_FMS_EASA_R0, issue dated 01.01.2009. Installation Instructions for the ATSO1 system for AS350 # doc. No. ATSO1_INST_EASA_R0 | 1. This STC provides approval for structural and electrical provisions only. Installation and operation of any equipment to be connected to the external antenna shall be covered by separate approval. 2. Prior to installation of this modificatio (...) | 03.03.2009 |
| 10017293 | ECMS AVIATION SYSTEMS GMBH | EASA.R.S.01566 | | | BELL 206A BELL 206A-1 BELL 206B BELL 206B-1 BELL 206L BELL 206L-1 BELL 206L-3 BELL 206L-4 | Installation of Bilfilar Cargo System | | Flight Manual Supplement: Doc. No. BCS01_FMS_EASA_R0, dated 06.04.2009 | 1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2) Th (...) | 08.06.2009 |
| 10028706 | ECMS AVIATION SYSTEMS GMBH | | | | BELL 212, 412 | Load Attachment Device BRD01 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval EASA CS 29 Amdt. 2 The certificated noise and/or emissions levels of the original produ (...) | Flight Manual Supplement for External Load Device BRD01 on Bell 212 and Bell 412, Doc. No. BRD01_FMS_EASA_R0, dated 01.12.2009 or later revisions of the above listed documents approved by EASA | None | 04.02.2010 |
| 10017166 | ECT INDUSTRIES | EASA.R.S.01274 | | | AS 350 B3 | Installation of ECT Industries DMS ARINC data recorder p/n REC-AR0-TT-00 | | ECT Industries definition file n° DD-0074-XX-82, revision 03, dated 20/10/2006, or later EASA approved revision. Installation: ECT Industries installation instructions n° IM-0078-XX-82-FR, revision 00, dated 28/11/2006, or later revision. Instructions (...) | | 04.04.2007 |
| 10017228 | ECT INDUSTRIES | EASA.R.S.01399 | REV. 4 | | AS355F1, AS355F2 AS355N, AS355NP | Installation of BRITE SAVER 2 Data Monitoring System. Revision 1 - covered a documentary change to STC revision 2 - covered torque and memory management minor change. Revision 3 - covered installation of ultrasonic air/ground switch, second possible locat (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The change under revision 4 complies with the requirements of FAR Part 27, amts 27-1 th (...) | ECT Industries AS355 Flight Manual Supplement RFMS-ARA-AS355-EN, second issue, revision 01 dated 09 February 2012 or later revisions of the above listed documents approved by EASA. Conformity Statement and Documentary Architecture for STC DMS AS355 -doc (...) | None | 15.05.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|------------------|----------------|----------------------|--|--|--|---|--|---------------|
| 10017248 | ECT INDUSTRIES | EASA.R.S.01446 | REV. 2 | | AS350 B3, EC 130 B4 | Installation of BRITE SAVER 2 Data Monitoring System (DMS) This STC is a Revision to an already EASA approved STC ref. 10017248, rev 1 dated 31/08/2009 to include Software Evolution and upgrade of CPU and analogical card of the data Monitoring System BRITE SAVER 2 | - AS 350 B3: FAR Part 27 Amendment 1 to 10 included- EC 130B4: JAR 27 First Issue dated September 6, 1993- The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this c (...) | BRITE SAVER 2 Amendment 2 Conformity Statement and Documentary Architecture CONFAD-0066-92-EN Issue Rev. 03 dated 01/08/2008 | None | 19.10.2009 |
| 10017272 | ECT INDUSTRIES | EASA.R.S.01494 | | | ROBINSON R22 ROBINSON R22 ALPHA ROBINSON R22 BETA ROBINSON R44 ROBINSON R44 II | DMS R22 & R44 - Brite Saver PE Usage Monitoring System for Robinson R22 and R44 Series Helicopters | | Installation, operation and maintenance of the DMS R22 & R44 Usage Monitoring System shall be in accordance with the following documentation: # Definition Document - DD-0004-08BE58-EN Issue 3 or later revision # Mounting Instructions - IM-KIT-PE- (...) | # Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. (...) | 13.01.2009 |
| 10027406 | ECT INDUSTRIES | P-EASA.R.S.01417 | REV. 3 | | EC 120B | Installation of ECT Industries DMS P/N REC-ARAO-TT-00 Rev. 1 adds a software change in the Data Monitoring System "BriteSaver2" and an RFM Supplement Rev. 2 adds a Usage Monitoring System for EC120 - "Brite Saver 2 amdt 4" | JAR 27 First Issue dated September 06, 1993 The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -ECT Industries Definition File no DD-0027-92-EN Rev. 12-ECT Industries Conformity Statement CONFAD-0096-92-EN Rev. 04-ECT Industries Installation Instruction no IM-KIT-ARA-EC 120B-EN Rev. 02-ECT Industries Operating Manual MUL-0031-92-EN Rev. 04-ECT (...) | None | 19.09.2011 |
| 10027594 | ECT INDUSTRIES | P-EASA.R.S.01487 | REV. 6 | | EC120B | Installation of Night Image Vision System kit, allowing night VFR flight with Night Vision Goggles Rev. 2 changes temperature limitation in Rotorcraft Flight Manual Rev. 3 incorporates minor changes to NVIS Shunt and LACUR Rev. 4 displays administrative (...) | JAR 27 First Issue dated September 06, 1993 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels o (...) | - Rotorcraft Flight Manual Supplement document RFMS-0010-09BE10-EN Revision 08, dated June 22, 2010- Maintenance Manual Supplement EC120 NHE NVIS Lighting System document MMS-57B-1000-000, revision 02, dated June 28, 2011 or later Revisions of the above (...) | Refer to applicable RFM for list of equipment mandatory for Night Vision Image System certification EC120B equipped with STC 10027594 Revision 4 | 19.09.2011 |
| 10029479 | ECT INDUSTRIES | P-EASA.R.S.01541 | | | AS355E, AS355F, AS355F1 AS355F2, AS355N, AS355NP | STC DTS AS 355 - Installation of a Tracking system. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS27 Change 1. The certificated noise and/ or emissions levels of the original products (...) | RFMS-DTS-355-EN revision 0 # Rotorcraft Flight Manual Supplement or later revisions of the above listed documents approved by EASA CONFAD-0021-053670-EN revision 3 dated 26 January 2010 # Conformity Statement: IM-0011-053670-EN revision 5 dated 9 Novemb (...) | None | 26.03.2010 |
| 10033164 | ECT INDUSTRIES | P-EASA.R.S.01505 | REV. 3 | | EC135 P1 (CPDS) EC135 P2 (CPDS), EC135 P2+ EC135 T1 (CPDS) EC135 T2 (CPDS), EC135 T2+ (...) | EC135 NVIS Lighting System Installation of a Night Vision Imaging Lighting system, to enable operation with Night Vision Goggles - Avionic type 1 - P/N: 60D-1000-000, into helicopters with CPDS and conventional flight/navigation instruments - Avionic type 2 (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 27 Amdt. 2 for the change The requirements for environmental protection and the associa (...) | - Eurocopter EC 135 NVIS Lighting RFM Supplement RFMS-0014-09BE11-EN Revision 07 or later revisions of the above listed documents approved by EASA - EC 135 NVIS Lighting System Conformity Statement and Documentary Architecture CONFAD-0001-09BE11-EN Revisi (...) | None | 27.01.2012 |
| 10043392 | ECT INDUSTRIES | P-EASA.R.S.01439 | | | BELL 206B BELL 206L BELL 206L-1 BELL 206L-3 BELL 206L-4 | BRITESAVER 2 DMS (Data Monitoring System) Installation. | | - IM-KIT-ARA-206-EN - BRITE SAVER 2 Bell 206 Mounting Instruction- MM1-0032-92-EN - BRITE SAVER 2 Maintenance Manual Level 1- MUL-0031-92-EN - BRITE SAVER 2 OPERATOR MANUAL | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The (...) | 25.06.2009 |
| 10032345 | EICHELSDOERFER GmbH | | | | RF5 | STC "Installation engine Rotax 912 and propeller HO-V352F-S1/S170FQ in the powered sailplane Fournier RF 5" | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | STC "Technische Mittelung 002-09-01" dated 27. April 2010. - Einbauanweisung 002-09-01, Einbau eines Rotax 912 A3 mit Propeller HO-V352F-S1/S170FQ in den Motorsegler RF 5. - Zeichnungsliste zum RF5-Rotax-Umbau- Flughandbuch für den Motorsegler Fourni (...) | Limited to Fournier RF 5, S/N 5012 | 28.10.2010 |
| 10040934 | EIRTECH AVIATION SERVICES LIMITED | | REV. 1 | | 737-600,-700,-800,-900,-900ER | Cabin Altitude and Take-off Configuration warnings and indications - Activation. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The applicant elected to comply with CS 25 Amendment 11 The requirements for environmen (...) | Eirtech Aviation Services Ltd Change ERT/CNG/153, Master Data List MDL/153-099 Issue 9 or later revisions of the above listed documents approved by EASA | This change is not applicable to an aircraft with High Altitude Landing Capability. | 26.10.2012 |
| 10045472 | EIRTECH AVIATION SERVICES LIMITED | | | | 737-300/-400/-500/-600/-700/ 737-800/-900/-900ER | Spectralux Dlink+ w/CPDLC installation (Link2000+ATN B1 compliant) in Boeing 737 aircraft | CS 25 Amendment 12 as amended by the following Special Conditions: F-01 High Intensity Radiated Fields F-03 Data Link Services for the Single European Sky The requirements for environmental protection and the associated certificated noise and/ or emissions (...) | Master Documents List MDL/282-050 Issue 04 Flight Manual Supplement FMS/282-002 Issue 01 Maintenance Manual Supplement MMS/282-012 Issue 01 Service Bulletin ERT/SB-162 Issue 01 or later revisions of the above listed documents approved by EASA | This CPDLC installation is limited to providing a supplementary means of communication; voice shall remain the primary means of communication. CPDLC shall only be used for routine data link exchanges during en-route operations in upper airspace. Prior to (...) | 27.06.2013 |
| 10046455 | EIRTECH AVIATION SERVICES LIMITED | | REV. 1 | | A318-111/-112/-121/-122 A319-111/-112/-113/-114 A319-115/-131/-132/-133 A320-111/-211/-212/-214 A320-215/-216/-231/-232 A320-233 A321-111/-112/-131 A321-211/-212/-213 A321-231/-232 | Installation of Spectralux Dlink+ Controller Pilot Data Link Communication (CPDLC) | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - Special Condition(s): F-3309-001 - "Data Link Services for the Single European Sky" - the following paragraph(s) at a lat (...) | Master Document List - MDL/158-099 Issue 5, dated 20/09/2013 Compliance Check List - CCL/158-099 Issue 5, date 30/08/2013 Declaration of Compliance - DOC/158 Issue 5, dated 20/09/2013 Flight Manual Supplement - FMS/158-007, Issue 3 dated 17/09/2013, o (...) | None | 20.09.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|-------------------|----------------|----------------------|--|--|---|---|--|---------------|
| 10016491 | EJM AEROSPACE SERVICES | EASA.IM.A.S.02853 | | FAA STC ST02534AT | BOEING 747-400 | Installation of a Matsushita Avionics Systems Corporation System 3000 In-Flight Entertainment System on a Singapore Airlines Boeing 747-400 in accordance with Master Drawing List MDL 10321, Rev. H, dated 29/04/2003, or later FAA approved revision. | | FAA approved Airplane Flight Manual Supplement, Document number 14526002, dated June 06, 2002 or later FAA approved revision. | This STC is approved only for the product configuration as defined in the approved design data. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.08.2008 |
| 10028392 | EJM AEROSPACE SERVICES | | | FAA STC ST02531AT | B747-400 | Installation of a Matsushita Avionics Systems Corporation System 3000 In-Flight Entertainment system on a Singapore Airlines Boeing 747-400 type aircraft in accordance with Master Drawing List MDL 10270, Rev. G, dated 06 June 2003 or later FAA approved revision. Validation of FAA STC ST02531AT | The Certification Basis for the original product remains applicable to this certificate/ approval | MAS Document Nr. 14526007 Rev E Instructions for Continued Airworthiness for MAS Corp System 3000 IFEMAS Document Nr. 14526003 Rev. IR Airplane Flight Manual Supplement for MAS Corp System 3000 IFE or later revisions of the above listed documents approv (...) | Not Applicable | 04.01.2010 |
| 10015335 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.00360 | | FAA STC ST01449CH | BOEING 757-200 | Dual GPS Antenna and Coax Cable (FAA STC ST01449CH) as per Master Data List ECS-200555 revision B dated June 11, 2003 or later FAA-approved revision. | | | 1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 2. If transfer of this Certificate is requested under Part 21A.116, the STC (...) | 07.12.2004 |
| 10015336 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.00361 | | FAA STC ST01839CH | BOEING 757-200 | Honeywell MK V Enhanced Ground Proximity Warning System (FAA STC ST01839CH) as per Master Data List ECS-201736 revision H dated 25 May 2004 or later FAA-approved revision. | | | 1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 2. If transfer of this Certificate is requested under Part 21A.116, the STC (...) | 07.12.2004 |
| 10015398 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.01101 | | ST02126CH | BOEING 757 | Installation of elementary & enhanced Mode S Transponders | | Electronic cable Specialists Master Data List ECS-202731 revision C dated March 4 2005 or later approved revision AFM Supplement as listed in Electronic Cable Specialists Master Data List ECS-202731 revision C dated March 4 2005 or later approved revision | | 26.07.2005 |
| 10015504 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.01267 | | ST01818CH | DC-9-21 DC-9-32, -33F, -34, -34F, -41 DC-9-51 DC-9-81 (MD-81) DC-9-82 (MD-82) DC-9-83 DC-9-87 (MD-87), -88, MD90-30 | Installation of a Honeywell Enhanced Ground Proximity System | | Electronic Cable Specialist Master Data List ECS-200489 Rev. AB dated 4 May 2005 Airplane Flight Manual Supplement as listed in Appendix A of Electronic Cable Specialists Master Data List ECS-200489 Rev. AB, limited to the models listed in the section "applicability" of this document. | None | 21.04.2006 |
| 10015561 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.01459 | | ST01404CH | DC-9-10 SERIES DC-9-20 SERIES DC-9-30 SERIES DC-9-40 SERIES DC-9-50 SERIES DC-9-81 (MD-81) DC-9-82 (CF8) DC-9-83 (MD-83) DC-9-87 (MD-87) | Installation of Universal Avionics system Corporation Multi-Function Display in accordance with Electronic Cable Specialists Master Data List ECS-2000416 rev. E dated 17.3.2003 | | Airplane Flight Manual Supplement, Electronic Cable Specialist Doc. No. ECS-200418 rev. A dated 19.9.2002 | | 17.02.2006 |
| 10015679 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.01660 | | ST01841CH | BOEING 747-100 BOEING 747-200C BOEING 747-200F BOEING 747-300 | Installation of GPS Antenna Provisions B747 Series in accordance with ST01841CH | | Data package containing the following documents, and additional supporting data: 1. Drawing 100-91347-1-NC Structural Provisions Installation GPS Antenna Left 2. Drawing 100-91347-2-NC Structural Provisions Installation GPS Antenna Right 3. (...) | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph 5 Documentation. Compatibility with other aircraft/engine configurations shall be determined by the installer. 2. The Boeing 747SR is not included in the EASA STC. | 31.07.2006 |
| 10015680 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.01661 | | ST01877CH | BOEING 747-100 BOEING 747-200B BOEING 747-200C BOEING 747-200F BOEING 747-300 | Installation of a Honeywell Ground proximity Warning System in accordance with FAA STC ST01877CH | | See item 5 of Approval Report n°: P-LVT-0035-2005 / EASA.IM.A.S.01661 dated 28th March 2006. | | 19.05.2006 |
| 10015682 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.01666 | | ST01923CH | LOCKHEED 1011-385-1 LOCKHEED 1011-385-1-15 | Installation of a Honeywell Enhanced Ground Proximity Warning System in accordance with ST01923CH | | Data package containing the following documents, and additional supporting data: 1. Certification Plan L1011. Installation of Honeywell International EGPWS with Terrain Display on Lockheed L1011 Series Aircraft (configuration 1 upgrade), Electronic (...) | 1. Airplane Flight Manual Supplement, Electronic Cable Specialists Doc. No. ECS-202314, Revision A, approved September 23, 2005, or later FAA approved revisions, are required on board the aircraft. 2. Airplane Flight Manual Supplement, Electronic Cable (...) | 02.08.2006 |
| 10015684 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.01668 | | ST01887CH | DC-9-32 DC-9-51 DC-9-81 (MD-81) DC-9-82 (CF8) DC-9-83 (MD-83) DC-9-87 (MD-87) MD-88 | Installation of GPS antenna provisions in accordance with FAA STC ST01887CH | | Electronic Cable Specialists Master Data List ECS-202152 Rev A dated 3rd August 2004 Instructions for Continued Airworthiness ECS Doc No ECS-202154 dated 9th January 2004 | The equipment for which these provisions are intended has not been certified. Additional EASA approval is required for the installation of this equipment | 13.03.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|-------------------|----------------|----------------------|--|--|---------------------|---|--|---------------|
| 10015692 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.01680 | REV 2 | FAA STC ST02162CH | BOEING 737-300 BOEING 737-400 BOEING 737-500 BOEING 737-600 BOEING 737-700 BOEING 737-800 | Installation of elementary or Enhanced surveillance Mode S transponders. | | 1. Electronic Cable Specialists Master Data List ECS-203150, Revision P, dated September 18, 2007; 2. Airplane Flight Manual Supplement as listed in Appendix A of Electronic Cable Specialists Master Data List No. ECS-203150, Revision P; 3. In (...) | 1. Only configurations 1, 2, 4, 5, 7, 8 and 9 as listed in the Master Data List ECS-203150, Revision P, are deemed to be approved; 2. The AFM Supplement as listed in Appendix A of Electronic Cable Specialists Master Data List ECS-203150, Revision P (...) | 30.07.2008 |
| 10015701 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.01689 | REV 1 | FAA STC ST02184CH | DC 10-30 SERIES DC-10-10 SERIES MD-11 | Installation of elementary and enhanced surveillance Mode S Transponders In MD-11; MD-11F; DC-10-10; DC-10-15; DC-10-30; DC 10-40; DC-10-10F; DC#10-30F; DC-10-40F in accordance with #Electronic Cables Specialists Master Data List ECS-203475#. | | 1. ECS-203482 Rev. N/C AFM Supplement for McDonnell Douglas MD-11 Series with Enhanced Mode-S Transponder Upgrade with flight ID from Transponder Control Panel.2. ECS-204353 Rev. N/C AFM Supplement for Douglas DC-10 Series Airplane with Mode-S (...) | 1. ECS-203482 Rev. N/C AFM Supplement for McDonnell Douglas MD11 Series with Enhanced Mode-S Transponder Upgrade with Flight ID from Transponder Control Panel.2. ECS 204353 Rev N/C AFM Supplement for Douglas DC-10 Series Airplane with Mode-S T (...) | 27.11.2007 |
| 10015799 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.01863 | REV 1 | FAA STC ST02007LA | BOEING 767-300 SERIES | Installation of Honeywell MCS-6000 SATCOM System | | FAA STC ST1193LA-T dated 2 March 2007 Master Data List ECS-203976 Revision B Flight Manual Supplement ECS-203980 | This installation must be installed in conjunction with Honeywell STC EASA.IM.A.S.02276 for the installation of the SATCOM High-Gain Antenna and Diplexer / Low Noise Amplifier (DLNA). | 08.03.2007 |
| 10015889 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.02008 | | FAA STC STC02075CH | AIRBUS A300 MODEL B2-1A AIRBUS A300 MODEL B2-1C AIRBUS A300 MODEL B2K-3C AIRBUS A300 MODEL B4-103 AIRBUS A300 MODEL B4-203 AIRBUS A300 MODEL B4-2C | Installation of Universal Avionics Systems Corporation Terrain Awareness and warning System (TAWS) | | Master Data List ECS-202603, revision G, dated 12 April 2006 or later approved revision | Aircraft Flight Manual Supplement ECS 202604, revision B, dated 11 October 2005 or later approved revision; instruction for continued airworthiness ECS 202604, revision B, dated 5 December 2005; if the holder agrees to permit another person to use this c (...) | 20.09.2006 |
| 10015992 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.02146 | | ST00694CH | DC-9-81 (MD-81) DC-9-82 (MD-82) DC-9-83 (MD-83) DC-9-87 (MD-87) | Installation of Dual Honeywell HT9100 GNSS GPS Navigation System. | | 2. Master Data List ECS-970034, Rev. R; 2. Summary Compliance Report ECS-206105, Rev. N/C; 3. AFM Supplement ECS-970047, Rev. B; 4. Instructions for Continued Airworthiness ECS-204077, Rev. A | 1. Limited to configuration III and IV.2. Compatibility of this design change with previously approved modifications must be determined by the installer.3. This STC is approved only for the product configuration as defined in the approved d (...) | 21.05.2007 |
| 10016069 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.02249 | | FAA STC ST02316CH | DC-9-80 SERIES | FAA STC ST02316CH - Elementary and Enhanced Mode S. | | ECS Master Data List ECS-203350 Rev. G dated July 7, 2006. | Limited to configuration 3 of ECS MDL ECS-203350 Rev. G dated July 7, 2006. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "#Associated Technical Documentat (...) | 28.02.2007 |
| 10016124 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.02317 | | FAA STC ST02303CH | BOEING 767-200, -300 SERIES | Installation of Enhanced Mode-S Transponder. | | Electronic Cable Specialists Master Data List, Document No. ECS-202463, Revision F, dated June 27, 2006 or later approved revision. | The Airplane Flight Manual Supplement as listed in Appendix A of Electronic Cable Specialists Master Data List, Document No. ECS-202463, Revision F, dated June 27, 2006, or later approved revision is required on board the modified aircraft. | 14.06.2007 |
| 10016386 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.02714 | | FAA STC SA01527CH | PILATUS PC-12 PILATUS PC-12/45 PILATUS PC-12/47 | Installation of a Honeywell KMH 880 Multi-Hazard Awareness System or a Honeywell KTA 870 Traffic Advisory System. | | FAA approved Electronic Cable Specialists Master Data List PC-12, PC-12/45, PC-12/47, Installation of IHAS, ECS-200836, Revision L, dated August 29, 2006, or later FAA approved revision. FAA approved Airplane Flight Manual Supplement as listed in Appendix A (...) | Limitations as per Master Data List PC-12, PC-12/45, PC-12/47, Installation of IHAS, ECS-200836 Revision L, dated August 29, 2006, or later FAA approved revision, Part III. Limitations | 10.03.2008 |
| 10016388 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.02716 | | FAA STC ST01479CH | BOEING 737-300-900 SERIES | Installation of a Smiths Industries Flight Management System Data Bus Portable Computing Device Recorder. | | Master Data List, Doc. No. ECS-200447, Revision K dated June 26, 2007 Airplane Flight Manual Supplement Doc. No. ECS-200675, Revision B, approved October 5, 2007 is required on board for MDL configuration 1, 2, and 3 Airplane Flight Manual Supplement (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 17.07.2008 |
| 10016507 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.02873 | | FAA STC ST02182CH | BOEING 747-100 BOEING 747-200B BOEING 747-200C BOEING 747-200F BOEING 747-300 BOEING 747-400 BOEING 747-400D BOEING 747-400F BOEING 747-SR | Installation of Elementary or Enhanced Surveillance Mode S Transponder law Electronic Cable Specialists Master Data List ECS-202128, rev. AB, 7 January 2008 or later approved revision. | | Electronic Cable Specialists AFM supplement as listed in Appendix A of Electronic Cable Specialists Master Data List ECS-202128, rev. AB, 7 January 2008 or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 20.10.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|---------------------|----------------|----------------------|---|---|---|--|---|---------------|
| 10016659 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.03120 | | FAA STC ST02612CH | AIRBUS A318-111 AIRBUS A318-112 AIRBUS A318-121 AIRBUS A318-122 AIRBUS A319-111 AIRBUS A319-112 AIRBUS A319-113 AIRBUS A319-114 AIRBUS A319-115 AIRBUS A319-131 AIRBUS A319-132 AIRBUS A319-133 AIRBUS A320-111 AIRBUS A320-211 AIRBUS A320-212 AIRBUS A320-214 AIRBUS A320-231 AIRBUS A320-232 AIRBUS A320-233 AIRBUS A321-111 AIRBUS A321-112 AIRBUS A321-131 AIRBUS A321-211 AIRBUS A321-212 AIRBUS A321-213 AIRBUS A321-231 AIRBUS A321-232 | Installation of AD Aerospace Cockpit Door Monitoring System | | # Master Data List, ref. doc. ECS-206333, Revision D, dated 25th June 2008 (*)# Airplane Flight Manual Supplement, ref. doc. ECS-206336, Revision N/C, approved 22nd July 2008 (*)# Instruction for Continued Airworthiness, ref. Doc.206328, Re (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 28.04.2009 |
| 10027156 | ELECTRONIC CABLE SPECIALISTS, Inc. | P-EASA.IM.A.S.01665 | | FAA STC ST00646CH | L1011-385-1-15, L1011-385-3 L1011-385-1, L1011-385-1-14 | Installation of a single or dual Honeywell/Trimble HT900 Global Navigation System, configurations 1, 2 or 3 in accordance with Electronic Cable Specialists Master Data List, ECS-960151, Rev. H dated January 8, 1999 or later FAA approved revisions. Validation of FAA STC ST00646CH | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The applicant has also shown compliance to Guidance Material AMC 20-5. | Airplane Flight Manual Supplement, document ECS-960150 Aircraft Maintenance Manual Supplement, document ECS-960169 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 10.09.2009 |
| 10027270 | ELECTRONIC CABLE SPECIALISTS, Inc. | P-EASA.IM.A.S.03122 | | FAA STC ST02637CH | 737-100, 737-200, 737-200C, 737-300, 737-400, 737-500, 737-600, 737-700, 737-800, 737-900, 737-900ER | Installation of ECS EZMount (TM) mechanical mounting assemblies | In accordance with TCDS The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Electronic Cable Specialists Documents: Master Data List ECS-206467 Rev. K dated July 31, 2009 AFM Supplement ECS-206471 Rev. A (applicable for EZMount (TM) Slider/AFM Supplement ECS-207424 Rev. A (applicable for EZMount (TM) Pivot) Instruction for Cont (...) | see limitations and conditions on STC FAA ST02637CH | 17.09.2009 |
| 10028835 | ELECTRONIC CABLE SPECIALISTS, Inc. | P-EASA.IM.A.S.03101 | | FAA STC ST02626CH | A318111/-112/-121/-122 A319-111/-112/-113/-114/-115 A319-131/-132/-133 A320-111/-211/-212/-214/ A320-214/-231/-232/ A321-111/-112/-131/-211/-212 A321-213/-231/-232 | Electronic Cable Specialists Master Data List, ref. doc. ECS-206919, Revision E, dated 9th September 2008 (*), Airplane Flight Manual Supplement, ref. doc. ECS-206922, Revision N/C, dated 29th September 2008 (*), Instructions for Continued Airworthiness (...) | Installation of an EZMount# Tilt Mechanical Mounting Assembly | | None | 02.06.2009 |
| 10033501 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.03101 | REV. 1 | FAA STC ST02626CH | A318-111, -112, -121, -122 A319-111, -112, -113, -114 A319-115, -131, -132, -133 A320-231, -232, -233 A321-111, -112, -131, -211 A321-212, -213, -231, -232 | Installation of an EZMount# Tilt Mechanical Mounting Assembly. Revision of the EASA STC EASA.IM.A.S.03101 to add Configuration 2 | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. 1. CRI H-01 "Enhanced Airworthiness Programme for Aerjets (...) Per FAA TCDS A22WE for parts not changed or affected by this change. The certification basis for the parts changed or affected by this change is the reference date of application, February 4, 2008 is shown on TCDS A22WE. In addition, the applicant has (...) | - FAA STC ST02626CH, amended 4th February 2010 - Electronic Cable Specialists Master Data List, ref. doc. ECS-206919, Revision H, dated 28th May 2010 - Airplane Flight Manual Supplement (for Config. 1), ref. doc. ECS-206922, Revision N/C, dated 29th Sep (...) | Not Applicable | 26.01.2011 |
| 10033754 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA.IM.A.S.03248 | REV. 1 | FAA STC ST02708CH | MD-11 | Installation of EZMount (TM) Arm mechanical mounting assemblies on lower frame of cockpit sliding window in accordance with Electronic Cable Specialists Master Data List ECS-207476 Rev. A, dated March 4, 2009, or later FAA approved revisions. Validation of FAA STC ST02708CH | Per FAA TCDS A22WE for parts not changed or affected by this change. The certification basis for the parts changed or affected by this change is the reference date of application, February 4, 2008 is shown on TCDS A22WE. In addition, the applicant has (...) | AFM Supplement as listed in Appendix A of ECS Master Data List ECS-207476 Rev. A | AFM Supplement as listed in Appendix A of Electronic Cable Specialists Master Data List ECS-207476, Rev. A, dated March 4, 2009 or later FAA approved revision, is required on board the modified aircraft. The equipment for which these provisions are intended (...) | 12.08.2013 |
| 10039706 | ELECTRONIC CABLE SPECIALISTS, Inc. | | | FAA STC ST02923CH | 737-600/-700/-800/-900/-900ER | Validation of FAA STC ST02923CH; EFB Installation of ECS EZMount Mechanical Mounting Assemblies | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List ECS-209276, Revision C, dated 26 January 2011, AFM Supplement ECS-209177, Initial release, dated 11 February 2011, EWIS Assessment ECS-209278, Revision A, dated 26 January 2011, ICA ECS-209223, Revision A, dated 7 January 2011 (...) | 1) This approval constitutes approval for the EZMount Mechanical Mounting Assemblies only. The capability of any system attached to the EZMount Mechanical Mounting Assembly requires additional EASA Approval. 2) The maximum allowed weight of any device (...) | |
| 10040901 | ELECTRONIC CABLE SPECIALISTS, Inc. | | | FAA STC ST02907CH | B747-400 B747-400F | Installation of ECS EZMount mechanical mounting assemblies in accordance with Electronic Cable Specialists, Inc. Master Data List ECS 209135, Revision H, dated November 29, 2011 or later approved revision. Validation of FAA STC ST02907CH | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. 14 CFR Part 25 Amendment 25-119 for the following requirements 25.301(a)(b); 25.303, 25 (...) | Master Data List ECS 209135, Revision H, Dated November 29, 2011 AFM Supplement ECS 208936 for Boeing 747-400 series Aircraft equipped with ECS EZMount (ECS-209136 Rev. NC Instructions for Continued Airworthiness ECS EZMount for Boeing 747-400, -400 (...) | The equipment for which these provisions are installed has not been certified. Additional approval is required for the installation of this equipment and must be evaluated to assure satisfactory compliance with applicable airworthiness standards. | 02.08.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|---------------------|----------------|----------------------|---|--|---|---|--|---------------|
| 10041328 | ELECTRONIC CABLE SPECIALISTS, Inc. | | | FAA STC ST02961CH | B737-600/-700/-800/-900/900ER | Validation of FAA STC ST02961CH - Approved configurations are the following:- Configuration 1: Replacement of Honeywell Weather Radar System RDR-4B with Honeywell Honeywell Weather Radar RDR-4000<->- Configuration 2: Software upgrade to Honeywell WR (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Configurations 1 and 2:- Master Data List ECS-208688, Revision G, dated 23 January 2012- ICA Baseline ECS-208689, Revision D, dated 7 October 2011- Configuration 1:- AFM Supplement ECS-208688, Initial release, dated 27 May 2011- EWIS/ICA ECS-208692, Revis (...) | None | 07.09.2012 |
| 10042532 | ELECTRONIC CABLE SPECIALISTS, Inc. | | | FAA STC ST02992CH | B737-600/-700/-800/-900/900ER | Validation of FAA STC ST02992CH - Installation of dual Electronic Flight Bag (EFB) System Provisions. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 "Enhanced Airworthiness Programme for Aeroplane Systems - ICAon EWIS" Issue 0 (...) | Master Data List (MDL), Reference ECS-209329, Revision K, FAA approved the 28 August 2012- Instructions for Continued Airworthiness (ICA), Reference ECS-209337<->-> Revision G, dated 22 August 2012- EWIS/ICA, Reference ECS-209751, Revision N/C, dat (...) | 1) The aircraft operator must perform the instructions for continued airworthiness and comply with the Airworthiness Limitation Items as per the Instructions for Continued Airworthiness (ICA), Reference ECS-209337, Revision G, dated 22 August 2012.2) Ad (...) | 06.12.2012 |
| 10042938 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA, I.M.A.S.01060 | | FAA STC ST01430CH | A300 B2-1A/B2K-3C/B2-1C/B2-203 A300 B4-2C/B4-103/B4-203 | Installation of universal Avionics Systems Corporation Electronic Horizontal Situation Indicators (EHSI) | | Master Data List ECS-200389, revision H, dated 04 November 2005 or later approved revision | 1. Aircraft Flight Manual Supplement ECS 200408 revision B, dated 15 November 2005 or later approved revision.2. Instruction for Continued Airworthiness ECS 200411, revision B dated 1 April 2005. If the holder agrees to permit another person to use this (...) | 20.09.2006 |
| 10043063 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA, I.M.A.S.02872 | | FAA STC ST01683CH | 747-200B, 747-200C, 747-200F 747-300, 747SR | Installation of Universal Avionics Systems Corporation Multi-Function Display in accordance with Electronic Cable Specialists Master Data List ECS-200663 Rev. G dated April 18, 2009 or later FAA approved revision. | | Airplane Flight Manual supplement as listed in Appendix A of Electronic Cable Specialists Master Data List ECS-200663 Rev. G, dated April 18, 2009 or later approved revision, is required on board the aircraft. | 1. Airplane Flight Manual supplement as listed in Appendix A of Electronic Cable Specialists Master Data List ECS-200663 Rev. G, dated April 18, 2009 or later approved revision, is required on board the aircraft.2. Prior to installation of this modifica (...) | 02.06.2009 |
| 10043149 | ELECTRONIC CABLE SPECIALISTS, Inc. | EASA, I.M.A.S.00404 | | FAA A16WE | B737-200, -300, -400, -500 | Installation of a Honeywell Enhanced Ground Proximity Warning System in accordance with Electronic Cable Specialists Master Data List ECS-201532, Revision B, dated March 17, 2003<->-> or later FAA approved revision. FAA STC ST01798CH amended on 24.n.20 (...) | | | 1. AFM Supplement as listed in Appendix A of ECS Master Data List Rev. T, ECS 201532 FAA Approved 23.11.2004. For configuration 13, GPS antenna installation must be accomplished prior with this modification in accordance with FAASTCST09490SC .2. This STC is (...) | 14.12.2004 |
| 10044506 | ELECTRONIC CABLE SPECIALISTS, Inc. | | | FAA STC ST03112CH | B747-400, B747-400F | Installation and activation of an Astronautics Corporation of America Electronics Class 3 Electronic Flight Bag (EFB) System consisting of an Electronic Unit (EU) and a Display Unit (DU). | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:- Equivalent Safety Finding(s): An equivalent level of safety findings has been made for the following regulation(s): CRI-E (...) | ECS-207709, rev. F, Electronic Cable Specialist Master Data List<->-> ECS-207678, rev. A, Project Compliance Report<->-> ECS-207718, rev. C, AFM Supplement<->-> ECS-207717, rev. A, Instructions for Continued Airworthiness<->-> ECS-210738, rev. N/C, EA (...) | Information displayed on the EFB is not to be used for primary navigation, and shall be used for reference only. This STC does not provide operational approval of the EFB system. Operational approval should follow the appropriate EASA approved guidance. T (...) | 27.05.2013 |
| 10044814 | ELECTRONIC CABLE SPECIALISTS, Inc. | | | FAA STC ST02949CH | B737-600/-700/-800/-900/900ER | Installation of a Rosemount Aerospace 8700E1 Class 3 Electronic Flight Bag (EFB) System. - Approved configurations are the following:- Configuration 1: Installation of Dual SDU electrical power with cross talk capabilities between the two units<->-> (...) | The Certification Basis for the original product and the EWIS CRI H-01 at issue 3 dated 20 August 2012. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification Plan, ECS-209047, Revision G, dated 27 April 2011- Project Compliance Report, ECS 209073, Revision D, dated 02 June 2011- EASA Certification Comparison for STC ST02949CH ECS-210104, Revision B, dated 04 January 2013- Master Data List, (...) | Limitations: 1) The aircraft operator must perform the instructions for continued airworthiness as per the document ECS-209658, Revision A, dated 25 April 2011. 2) Information displayed on the Electronic Flight Bag is not to be used for primary navigation (...) | 07.05.2013 |
| 10044973 | ELECTRONIC CABLE SPECIALISTS, Inc. | | | FAA STC ST03006CH | 737-600, 737-700, 737-800, 737-900, 737-900ER. | Retrofit installation of a Dual CMC Electronics Class 3 Electronic Flight Bag (EFB) System. | The Certification Basis for the original product and: the EWIS CRI H-01 at issue 3 dated 20 August 2012<->-> The Lithium Battery CRI F-01 at issue 3 dated 2 April 2013. The certificated noise and/or emissions levels of the original product are un (...) | Certification Plan, ECS-209314, Revision E, dated 8 July 2011- Project Compliance Report, ECS 209313, Revision D, dated 07 October 2011- EASA Certification Comparison for STC ST03006CH, ECS-210121, Initial release, dated 23 May 2012- Master Data List, E (...) | 1) The Operating System (OS) for which this Class 3 EFB system is intended has not been approved. Additional EASA approval is required for the installation of any Operating System and this must be evaluated to assure satisfactory compliance with the ap (...) | 23.05.2013 |
| 10045157 | ELECTRONIC CABLE SPECIALISTS, Inc. | | | FAA STC ST03007CH | B737-600/-700/-800/-900/900ER | Validation of FAA STC ST03007CH line 1 installation of a Dual CMC Electronics Class 3 Electronic Flight Bag (EFB) System. | The Certification Basis for the original product and: the EWIS CRI H-01 at issue 3 dated 20 August 2012<->-> The Lithium Battery CRI F-01 at issue 3 dated 2 April 2013. The certificated noise and/or emissions levels of the original product are un (...) | Certification Plan, ECS-208298, Revision G, dated 11 January 2012- Project Compliance Report, ECS 208299, Revision B, dated 19 January 2012- EASA Certification Comparison for STC ST03007CH, ECS-210256, Initial release, dated 23 May 2012- Master Data List, (...) | 1) The Operating System (OS) for which this Class 3 EFB system is intended has not been approved. Additional EASA approval is required for the installation of any Operating System and this must be evaluated to assure satisfactory compliance with the ap (...) | 04.06.2013 |
| 10013942 | ELECTRONIQUE MAINTENANCE AVIATION | EASA, A.S.00378 | REV. 1 | | FAIRCHILD SWEARINGEN SA 226-T(| Installation of VHF 8.33 KHz/VOR/ILS/GPS/STORMSCOPE/VHF 150 MHz/UHF/ICS Passengers according to E.M.A. Modification Approval Sheet (FAM) No EMA/03/002/SA226Series/VHF. GPS BRNAV STORM, UHF dated 22 December 2003. - Installation of GARMIN GNS 530A (...) | | 03/002/SA226Series/...R1- Flight Manual Supplement- Air flight test by CEV ISTRS | 1. This modification is applicable only to SA 226 T(B) s/n T298 F-GLPT.2. EASA Approved Airplane Flight Manual Supplements "Supplément au manuel de vol GPS BRNAV" are required.3. This STC is approved only for the product configuration as de (...) | 05.02.2009 |
| 10013968 | ELECTRONIQUE MAINTENANCE AVIATION | EASA, A.S.00738 | REV. 1 | | FAIRCHILD SWEARINGEN SA 226-T(| Installation of GPS BRNAV, EGPWS, TAS, Stormscope, Radar, MFD> Garmin GNS 530 VHF, NAV, GPS, Honeywell RDR 2000 Radar, BFGoodrich WX 500 Stormscope and Avidyne EX 500 MDF installation- BF Goodrich Sky 899 TAS and Honeywell KGP 560 EGPWS provision | | AFM Supplement:#Supplément au Manuel de Vol GPS BRNAV EMA/05/002/SA226/GPS, BRNAV, EGPWS, TAS, STORM, RDR, MFD Rev. 1 Avion No. T312, F-GRNT# | 1. This STC is limited to MSN T312 F-GRNT2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 20.02.2006 |
| 10014112 | ELECTRONIQUE MAINTENANCE AVIATION | EASA, A.S.01217 | | | BEECH C90 BEECH E90 | Allied Signal KT73 ATC Mode S, Kannad 406AF ELT 406 and 2 IDC 570 altimeters installation. | | EMA/05/002/BEECH90/ATC.ELT Revision 1 | This STC is limited to Beech C90 MSN L1667 & Beech E90 MSN LW 278. | 13.06.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|-------------------|----------------|--------------------------------|---|---|---------------------|--|--|---------------|
| 10014251 | ELECTRONIQUE MAINTENANCE AVIATION | EASA.A.S.01473 | | | BEECH 200 | 406 MHz ELT installation | | Airplane Flight Manual supplement No EMA04/002/BE200/ELT406, rev 1 of July 2005. | Modification applicable on s/n BB616 and BB202 only. | 05.08.2005 |
| 10014341 | ELECTRONIQUE MAINTENANCE AVIATION | EASA.A.S.01601 | | | FAIRCHILD SWEARINGEN SA 226T | ATC Mode S, ELT 406 MHz, Altimeter, Turn and Bank annunciator installation | | - AFM Supplement "Supplement au Manuel de Vol RBOA", STC No. EMA/02/003/SA226Serie/ATC.ELT.ALTI.T/B | | 30.01.2006 |
| 10014747 | ELECTRONIQUE MAINTENANCE AVIATION | EASA.A.S.02930 | | | FAIRCHILD SWEARINGEN SA 226T | Garmin GNS 530A VHF 8.33KHZ, Honeywell KY 196B VHF 8.33 KHZ, BF Goodrich TAS SKY 899 Installation | | Airplane Flight Manual Supplement ref. T321-899 revision IR, dated August 19, 2003 or later approved revision. | This STC is limited to the SA226 S/N T312 | 26.02.2007 |
| 10014812 | ELECTRONIQUE MAINTENANCE AVIATION | EASA.A.S.03019 | | | | Installation of Avionic Upgrade according EMA/07/001/SA226/VHF8.33-ATC-MFD-RDR | | Master Doc. List. EMA/07/001/SA226/VHF8.33-ATC-MFD-RDR | | 30.04.2007 |
| 10014834 | ELECTRONIQUE MAINTENANCE AVIATION | EASA.A.S.03050 | | | BEECH 200 | Installation of Garmin GTX330D Mode S Transponder. | | Installation in accordance with EMA STC Data List Doc-No EMA/07/001/BE200/ATC Mode S, Rev 1, or later approved Revisions To be operated in accordance with EISA Airplane Flight Manual Supplement Doc-No. 05 AFMS S128, Rev 0, or later approved Revisions T (...) | | 21.04.2008 |
| 10015045 | ELECTRONIQUE MAINTENANCE AVIATION | EASA.A.S.03361 | | | BEECH 200 | Installation of new COM-NAV-GPS with BRNAV capacity and coupled to AP. | | EMA/07/003/BE200Serie/VHF/NAV-GPS, dated November 2007, revision 0 AFM Supplement EMA/07/003/BE200Serie/VHF/NAV/GPS. | AFM Supplement must be used when the application is applied. Modification limited to s/n BB616. | 04.08.2008 |
| 10017146 | ELECTRONIQUE MAINTENANCE AVIATION | EASA.R.S.01249 | | | EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 EUROCOPTER FRANCE AS 355N | Installation of SAFE FLIGHT Power Line Detector p/n 7101-5, in accordance with EMA STC file n° 06/001/AS355Serie/PLD revision 0, dated January 2006, or later EASA approved revision. | | Installation and Continued Airworthiness SAFE FLIGHT installation manual ref. R2925 Operation: EMA RFM supplement ref. 06/001/AS355Serie/PLD revision 0, dated January 2006 or later EASA approved revision. | | 27.02.2007 |
| 10016097 | ELISEN TECHNOLOGIES | EASA.IM.A.S.02286 | | TCCA STC SA07-9 | BOMBARDIER CL-600-2B19 | Iridium Antenna Installation | | Transport Canada STC SA07-9, Elisen Technologies Inc. Master Data List MDL-29R2-639, Revision NC, dated 21 January 2007 or later Transport Canada approved revision. | | 31.05.2007 |
| 10016098 | ELISEN TECHNOLOGIES | EASA.IM.A.S.02287 | | TCCA STC SA07-3 | BOMBARDIER CL-600-2B19 | Aft Equipment Bay Steps | | Transport Canada STC No. SA07-3, Elisen Technologies Inc. Master Data List MDL-29R2-647, Revision NC, dated 21 January 2007 or later Transport Canada approved revision. | | 31.05.2007 |
| 10016099 | ELISEN TECHNOLOGIES | EASA.IM.A.S.02288 | | TCCA STC SA07-4 | BOMBARDIER CL-600-2B19 | Ladder Stowage | | Transport Canada STC SA07-4, Elisen Technologies Inc. Master Data List MDL-29R2-648, Revision NC, dated 21 January 2007, or later Transport Canada approved revision. | | 31.05.2007 |
| 10016100 | ELISEN TECHNOLOGIES | EASA.IM.A.S.02289 | | TCCA STC SA07-5 | BOMBARDIER CL-600-2B19 | Towbar Head Stowage | | Transport Canada STC SA07-5, Elisen Technologies Inc. Master Data List MDL-29R2-649, Revision NC, dated 21 January 2007, or later Transport Canada approved revision. | | 31.05.2007 |
| 10016101 | ELISEN TECHNOLOGIES | EASA.IM.A.S.02290 | | TCCA STC SA07-6 | BOMBARDIER CL-600-2B19 | Main wheel stowage | | Transport Canada STC SA07-6, Elisen Technologies Inc. Master Data List MDL-29R2-653, Revision NC, dated 21 January 2007, or later Transport Canada approved revision. | | 31.05.2007 |
| 10016168 | ELISEN TECHNOLOGIES | EASA.IM.A.S.02395 | | TCCA STC SA07-20 | BOMBARDIER CL-600-2B19 | Miscellaneous Storage Box, Project No. 29R2-707 | | MDL-29R2-707 rev NC or later approved revision. Instruction for continued airworthiness ICA-29RD-707 revision NC or later approved revision. | | 03.07.2007 |
| 10016169 | ELISEN TECHNOLOGIES | EASA.IM.A.S.02396 | | TCCA STC SA07-21 | BOMBARDIER CL-600-2B19 | Towbar Stowage, Project No. 29R2-708. | | MDL-29R2-708 revision NC or later approved revision. Instructions for continued Airworthiness ICA-29R2-708 revision NC or later approved revision. | | 03.07.2007 |
| 10016170 | ELISEN TECHNOLOGIES | EASA.IM.A.S.02397 | | TCCA STC SA07-22 | BOMBARDIER CL-600-2B19 | Equipment Bay Forward Catwalk, Project No. 29R2-709. | | MDL-29R2-709 revision NC or later approved revision. Instructions for continued Airworthiness ICA-29R2-709 Revision NC or later approved revision. | | 03.07.2007 |
| 10016171 | ELISEN TECHNOLOGIES | EASA.IM.A.S.02398 | | TCCA STC SA07-23 | BOMBARDIER CL-600-2B19 | Equipment Bay Stowage Box, Project No. 29R2-710. | | MDL-29R2-710 revision NC or later approved revision. Instructions for continued airworthiness ICA-29R2-710 revision NC or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 04.07.2007 |
| 10016310 | ELISEN TECHNOLOGIES | EASA.IM.A.S.02612 | | TRANSPORT CANADA STC NO. SA07- | BOMBARDIER CL-600-2B19 | Lockable Fuels Caps. | | Elisen Technologies Inc. Master Data List MDL-29R2-70C revision NC dated 18 April 2007. | | 21.11.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------|-------------------|----------------|----------------------|---|--|---------------------|---|--|---------------|
| 10015349 | ELLIOTT AVIATION | EASA.IM.A.S.00503 | REV. 1 | FAA STC SA01352CH | BEECH 200 BEECH 200C BEECH 200CT BEECH 200T BEECH 300 BEECH 300LW BEECH B200 BEECH B200C BEECH B200CT BEECH B200T BEECH B300 BEECH B300C | Installation of Dual TDR-94D Transponders with Flight ID in B200/B200C/B300/B300C Aircraft with Pro Line 21 Avionics, in accordance with Configuration 3 of Elliott Aviation Master Data List, No 2026-576DL, Revision 2 or later approved revision. | | - Airplane Flight Manual Supplement, Elliott Aviation Doc. No. FS2026-344, Revision 0 or later EASA approved revision, is required on board the aircraft | 1. Limited to Aircraft equipped with Raytheon Aircraft Company "Collins Pro Line 21 Avionics Suite" major change 2. Limited to Configuration 3 of Elliott Aviation FAA STC SA01352CH3. This STC is approved only for the product configuration as defined in the approved design data referred to in | 06.03.2006 |
| 10015683 | ELLIOTT AVIATION | EASA.IM.A.S.01667 | | FAA STC ST02007CH | BEECH 400 BEECH MU-300 BEECH MU-300-10 | Certification for Reduced Vertical Separation Minimum (RVSM) operational capability and installation of Honeywell AZ-252 Air Data Computer and AM-250 Altimeter with optional BA-250 Altitude Indicator and AL-800 Altitude Preselector | | - FAA STC ST02007C: Elliott Aviation Master Data List No. 2022-462DL Rev. 3 dated July 07, 2004 or later EASA approved Revisions- Airplane Flight Manual Supplement, Doc. No. FS2022-462 Rev. 1 dated July 20, 2004 or later EASA approved Revisions- Ins (...) | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer.2. Certificati (...) | 24.01.2006 |
| 10015693 | ELLIOTT AVIATION | EASA.IM.A.S.01681 | | FAA STC ST01486CH | BEECH 400 BEECH MU-300 BEECH MU-300-10 | Installation of a Honeywell Mark VI/VIII EGPWS | | FAA STC ST01486CElliott Aviation Master Data List No. 2024-571DL Rev. 4 dated May 08, 2006 or later EASA approved RevisionsAirplane Flight Manual Supplement for Honeywell MK VI & VIII EGPWS, eport No. FS2024-571 Rev. 2 dated June 02, 2006 or later E (...) | Validation Basis of the Change: FAR 25, additionally JAA Leaflet No. 12 (TAWS) | 19.07.2006 |
| 10015716 | ELLIOTT AVIATION | EASA.IM.A.S.01713 | | ST02194CH | FALCON 10 | ST02194CH: Installation of UNS EFI-890R EFIS | | -MDL-2463-000 Rev. E dtd. 2.Jan 2007 or later approvedrevision--AFM supplement(AFM) as per MDL-2463-000 Rev. E--AFMSFMS-2651-004 Rev.A - <no date> or later approvedrevision OR--AFMS FMS-2463-000 Rev.B dtd. 1.Dec 2006 or laterapproved revision-In (...) | Certification basis iaw CRI A1 I.M.A.S.01713 | 13.02.2007 |
| 10015721 | ELLIOTT AVIATION | EASA.IM.A.S.01720 | | FAA STC ST01666CH | FALCON 10 | ST01666CH: Installation of a Rockwell Collins TCAS 4000Traffic Alert and Collision Avoidance System (TCAS II) | | -MDL-2010-576DL Rev. 5 dtd. 20.October 2006 or later approvedrevision--AFM supplement(AFM) / Instructions for ContinuedAirworthiness(CAW) as per MDL-2010-576DL Rev. 5--AFMS FMS-2651-002 Rev.B dtd. 20.Oct 2006 or later approvedrevisionOR--AFMS F (...) | 1. Certification basis iaw CRI A1 I.M.A.S.01720 and TGL Nr. 8 | 13.02.2007 |
| 10015757 | ELLIOTT AVIATION | EASA.IM.A.S.01791 | | FAA STC SA02206CH | BEECH 200 BEECH 200C BEECH 200CT BEECH 200T BEECH 300 BEECH 300C | Installation of Honeywell Digital Flight Data Recorder System (DFDR) | | In accordance with LJSC Master Document List No. MDL5004-31290, Rev. H, dated November 7, 2005, or later approved revisions.FAA Approved Aircraft Flight Manual Supplement LJSC Doc. No. FMS5004-3129, Rev IR, dated November 3, 2005 or later approved r (...) | Limited to the aircraft types listed above with Proline 21 Avionics Suite installed. | 03.03.2006 |
| 10015894 | ELLIOTT AVIATION | EASA.IM.A.S.02014 | | FAA STC SA02264CH | BEECH 300 BEECH 300LW BEECH B200 BEECH B200C BEECH B200CT BEECH B300 BEECH B300C | Airworthiness certification of Reduced Vertical separationMinimum (RVSM) and installation of Air Data Systems | | Installation in accordance with Elliott Aviation Master Drawing List, Report No MDL-2468-000, Revision C, dated Amrch 16, 2006 or later approved revisionOperation in accordance with FAA Approved Airplane Flight Manual Supplement, Elliott Aviation Re (...) | None | 09.11.2006 |
| 10015976 | ELLIOTT AVIATION | EASA.IM.A.S.02128 | | FAA ST5990CH-T | FALCON 10 | ST02333CH: Installation of a Universal Avionics SystemsCorporation(UASC) UNS-1K-VL FLIGHT MANAGERMENTSYSYSTEM (FMS); Dual Universal Avionics Radio ControlUnits (RCU), and/or dual Rockwell Collins VIR-32Navigation Receiver | | -MDL-2651-000 Rev.C dtd 19.Oct.2006--AFM supplement(AFM) /Instructions for ContinuedAirworthiness(CAW) as per MDL-2651-000 Rev.C--AFMS FMS-2651-000 Rev. IR dtd.18. Jul 2006 or later approvedrevision OR--AFMS FMS-2651-001 Rev. A dtd.19.Oct 2006 (...) | 1. Certification basis FAR 25 Amdt 117 | 13.02.2007 |
| 10015978 | ELLIOTT AVIATION | EASA.IM.A.S.02130 | | SA01216CH | BEECH 200 BEECH 200C BEECH 200CT BEECH 300 BEECH 300LW BEECH B200 BEECH B200C BEECH B200CT BEECH B200T BEECH B300 BEECH B300C | Installation of Honeywell (Allied Signal) MK-VI Enhanced Ground Proximity Warning System, Universal Avionics Systems Corporation Terrain Awareness Warning System and Universal Avionics Systems Multifunction Display | | Elliott Aviation Master Document List 2007-571 MDL Rev 2, dated May 4, 2002 or later FAA approved revision. | 1. Compatibility of this design change with previously approved modifications must be determined by the installer.2. Airplane Flight Manual Supplement, Elliott Aviation Report No. EGPWS-FMS Rev 1, FAA approved August 23, 2000, or later FAAapproved re (...) | 24.08.2006 |
| 10016252 | ELLIOTT AVIATION | EASA.IM.A.S.02529 | | FAA STC SA02447CH | BEECH 200 BEECH 200C BEECH 200CT BEECH 200T BEECH 300 BEECH 300C | Installation of a Rockwell Collins IFIS-5000 Integrated Flight Information system. | | Installation must be in accordance with Elliott Aviation Master Data List No. MDL-2707-000 Rev. A, dated 26.June.07 or later Approved Revisions.Operation must be in accordance with Elliott Aviation Flight Manual Supplement FMS-2707-00 Rev.IR dated 0 (...) | | 27.02.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|---------------------|----------------|-------------------------|---|---|---|---|--|---------------|
| 10016254 | ELLIOTT AVIATION | EASA.IM.A.S.02532 | | FAA STC ST02431CH | FALCON 10 | EASA Validation of FAA STC ST02431CHEASA Validation is limited to Provisions for Installation of a Universal Avionics Application Server Unit (ASU) on Falcon 10 | | Installation in accordance with: # Elliott Aviation Master Data List, Report No. MDL-2710-000 Revision A, dated 15th May 2007, and with# Elliott Aviation drawing No. 300-2710-E0455-001 Rev. IR, dated 08th May 2008 for the deactivation of the ASU (...) | The aircraft must be equipped with Universal Avionics EFI-890R Electronic Flight Instrument System (EFIS) as installed by Elliott Avionics STC EASA.IM.A.S.01713. This EASA Approval is limited to provisions for installation only with the ASU System (...) | 13.05.2008 |
| 10016474 | ELLIOTT AVIATION | EASA.IM.A.S.02822 | | FAA STC ST02548CH | FALCON 10 | EASA Validation of FAA STC ST02548CHInstallation of an International Communications Group ICS-120 Iridium Satellite Telephone System | | Installation in accordance with Elliott Aviation Master Data List, Report No. MDL-2837-111 Revision B, dated 06th March2008,FAA AFM Supplement Elliott Aviation report No. FMS-2837-111; Revision IR, approved 14th March 2008 is required on board the a (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 26.05.2008 |
| 10016475 | ELLIOTT AVIATION | EASA.IM.A.S.02823 | | FAA STC ST02547CH | FALCON 10 | EASA Validation of FAA STC ST02547CHInstallation of a Universal Avionics Systems Corporation UL-701 UniLink System | | Installation in accordance with Elliott Aviation Master Data List, Report No. MDL-2836-114 Revision D, dated 23rd April2008, or later revisions of this document approved by EASA in accordance with EASA ED Decision 2004/04/CF (or i.a.w. subsequent re (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 26.05.2008 |
| 10029872 | ELLIOTT AVIATION | | | FAA STC STC02784CH | BEECHJET 400/400A | Installation of Honeywell Digital Flight Data Recorder/Validation of FAA STC STC02784CH | The Certification Basis for the original product remains applicable to this certificate/ approval | Elliott Aviation Master Data List, Report No. MDL-3001-573, Rev. B,dated December 16, 2009-FAA Approved Flight Manual Supplement, Elliott Aviation Report No.FMS-3001-573, Revision IR, approved December 14, 2009-Instructions for Continued Airworthin (...) | None | 28.04.2010 |
| 10029873 | ELLIOTT AVIATION | | | FAA STC STC02783CH | BEECHJET 400 | Installation of an Artex C406-2 Emergency Locator Transmitter (ELT)/Validation of FAA STC STC02783CH | The Certification Basis for the original product remains applicable to this certificate/ approval | Elliott Aviation Master Data List, Report No. MDL-3010-578, Rev. A,dated November 30, 2009-FAA Approved Flight Manual Supplement, Elliott Aviation Report No.FMS-3010-578, Revision IR, approved December 2, 2009-Instructions for Continued Airworthine (...) | None | 28.04.2010 |
| 10015712 | ELLIOTT AVIATION TECHNICAL PRODUCTS | EASA.IM.A.S.01709 | | FAA STC ST00875WI | FALCON 10 | ST00875WI: Installation of Universal TAWS with optionalUniversal MFD-640 and Allied Signal MKIVV EGPW with windshear | | Master Data List MDL 2011-571DL Rev. 5 dtd. 2.Jan07--AFM supplement (AFMS) / Instructions for ContinuedAirworthiness(CAW) as per MDL 2011-571DL--AFMS FS2011-571, Rev. 1, dtd. 12.Oct 2006 or later approved revision OR--AFMS FS2011-454, Rev. 1, dtd. (...) | Certification basis iaw CRI A1 I.M.A.S.01709 | 13.02.2007 |
| 10015890 | EMBRAER - EMPRESA BRASILEIRA | EASA.IM.A.S.02009 | | CTA STC2006S02-09 | EMBRAER EMB-135BJ | CTA STC2006S02-09 "Installation of a 30' LCD monitor and Playstation Entertainment System iaw Avionic Services master drawing list 135-23-001 Rev. D dtd. 28. Nov. 05 or later approved revision | | -- | 1. Limited to EMB-135BJ SN 145000802. Instructions for Continued Airworthiness Report No. 135-23-0008 Rev. 0 dtd. 9.Sep. 05 or later accepted revision.2. This STC is approved only for the product configuration as defined in the approved design (...) | 23.01.2007 |
| 10027373 | EMBRAER - EMPRESA BRASILEIRA | P-EASA.IM.A.S.03273 | | ANAC STC 2009S05-14 | EMB-500 | Installation of a Honeywell HF Communications System, ModelKHF-1050 andoptionally the Avtech Corporation SELCAL decoder, Model CSD-714. | CS-23 | Installation in accordance with Master Document List No TR0116, Rev H,dated 11 August 2009 or later approved revisions.Operation in accordance with Embraer EASA Airplane Flight ManualSupplement (AFMS) document No H.02-2921-0/ANAC/2009, RevIR, dated (...) | Required Markings and Placards:- In close proximity to the 5A breaker in panel LH circuit breaker "HFPWR".In close proximity to the 30A breaker in circuit breaker of LPDU "HFAMPL", (if applicable) In close proximity to the 5A breaker in panel LH circ (...) | 25.09.2009 |
| 10038234 | EMBRAER GPX Ltda. (PC 722) | | REV. 1 | | EMB-135BJL/ER EMB-145 EMB-145ER/EU/EP/MP/LR/LU/MK | EASA Validation of ANAC STC 2010S10-18. Electronic Flight Bag (EFB)Installation.Description as per:ESCMSC002-Rev A, dated 17th January 2012.Master Documents List ESCCCD003-Rev B, dated 17th January 2012. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | ESCMSC002-Rev A, dated 17th January 2012.Master Documents List ESCCCD003-Rev B, dated 17th January 2012.Flight Manual Supplement No.: H.02-3489-0/ANAC/2010, dated 25th March2011.or later revisions of the above listed documents approved by EASA inacc (...) | None | 01.02.2012 |
| 10038291 | EMBRAER GPX Ltda. (PC 722) | | | | EMB-135BJ | EASA validation of ANAC STC N°2011S09-02Installation of a cabinet and a 17" video monitor on EMB-135BJ aircraft | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | ANAC STC N° 2011S09-02Master Document List N° ESCCCD015, Rev. F, dated 01 Feb. 2012Airplane Flight Manual Supplement doc. N° H.02-3679/2011/ANAC Rev. Adated 01 Feb. 2012or later revisions of the above listed documents approved by EASA inaccordance w (...) | None | 03.02.2012 |
| 10039722 | EMBRAER GPX Ltda. (PC 722) | | | ANAC STC NO. 2009S10-20 | EMB-135BJ | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/or emissions levels of the original product areunchanged and remain applicable to this certificate/appoval. | EASA Validation of ANAC STC No. 2009S10-20, issued 20/10/2009.Installation of Erffite Refrigerator.Description as per:MASTER DOCUMENT LIST ref. TR0152 issue E, dated 18/04/2012 | MASTER DOCUMENT LIST ref. TR0152 issue E, dated 18/04/2012or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/03/CF (or subsequent revisions of this decision) | none | 18.05.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|-------------------|----------------|--------------------------------|--|--|--|--|--|---------------|
| 10031622 | EMBRAER S.A. | EASA.IM.A.S.03239 | | ANAC N°2009SOS-07 | EMB-500 | Installation of a Rockwell Collins Automatic Direction Finder ADF-462model | | Installation in accordance with Master Document List No TR0083, Rev J, dated 14 May 2009 or later approved revisions. Operation in accordance with Embraer EASA Airplane Flight Manual Supplement (AFMS) for Rockwell Collins ADF-462 ADF system, Rev IR, da (...) | 1. Required Markings and Placards:..In close proximity to the 5A ADF breaker in breaker panel placard "ADF".2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other pr (...) | 01.09.2009 |
| 10015319 | EMPIRE AIRLINES | EASA.IM.A.S.00260 | | FAA STC ST01444SE | ATR 42/72 | Installation of B/E Aerospace Sweep-On Fullface Crew Oxygen Mask System on the flight deck of ATR 42/72 series airplanes in accordance with Empire Airlines Document Number 3510, Revision Original dated May 25, 2004, or later FAA-approved revision. Th (...) | | | 1. Incorporation of Supplemental Type Certificate (STC) ST01189W1, passenger to all-cargo Class E cargo compartment conversion, must be accomplished prior to or concurrent with this STC.2. A copy of this Certificate and Empire Airlines Document Number 3510, Revision Original, dated May 25, 200 | 07.10.2004 |
| 10032105 | EMS AVIATION, Inc. | | REV. 1 | FAA STC SR03333AT | S-76A, S-76B, S-76C | Installation of the Sky Connect Satellite Phone and Tracking System. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Icarus Instruments Inc. Installation Drawing List 76-A2008-091-1, RevIR, dated June 6, 2008- Installation instructions 76-A2008-091-3, Rev A, dated August 8, 2008- EASA RFM Supplement RFM-A2005-091, Rev IR, dated 9/22/2010- ICA Document N° A2005-0 (...) | None | 21.09.2011 |
| 10038432 | EMS AVIATION, Inc. | | | FAA STC SR09583RC | EC135 P1/P2/P2+/T1/T2/T2+ | Installation of the Sky Connect Forte Iridium Satellite Phone and /or automated reporting system (FAA STC SR09583RC) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement RFMS-2009-300 Rev A, dated 11/17/2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)-Master Drawing List MD (...) | None | 16.02.2012 |
| 10041017 | EMS AVIATION, Inc. | | | FAA STC NO SR02569NY | AS 350 B, AS 350 B1, AS 350 B2, AS 350 B3, AS 350 BA, AS 350 D | Installation of Sky Connect Tracking System as Master Drawing List MDL07094. This STC is the validation of FAA STC SR02569NY (last reissued April 15, 2009) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Definition and installation, Inspection and Maintenance: Avionics Design Services Ltd. Master Drawing List MDL07094 Revision C Issue 1 dated July 02, 2008 Avionics Design Services Ltd. Instructions for Continued Airworthiness ICA07094-1 Rev. B dated Janu (...) | This STC is not applicable to AS350B3 model helicopters with modification OP4305 (Arriel 2D engine installation) incorporated. | 15.08.2012 |
| 10045559 | EMS AVIATION, Inc. | | | FAA STC ST02696NY | GULFSTREAM G-IV | Installation of Sky Connect Forte Iridium Satellite phone tracking system with Network Gateway | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -FAA STC FAA ST02696NY, reissued 18th November 2010-Sky Connect LLC Certification Plan, Doc. No. CP-2009-400, Revision D, dated 22nd March 2010-Sky Connect LLC Installation Drawing List Doc. No. GIV-2009-100-1, Revision A, dated 7th October 2009-Sky Co (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 04.07.2013 |
| 10016107 | EMS TECHNOLOGIES CANADA LTD | EASA.IM.A.S.02297 | | TCCA STC SA05-12, ISSUE 3, DAT | | Installation of EMS HSD-64/-128/-X/-400/-400/-HD-710 High Speed Data Transceiver | | EMS HSD-64/-128/-X/-400/-HD-710 High Speed Data Transceiver, Engineering Report RPT04021 by Avionic Services Ltd | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "#Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer | 27.06.2007 |
| 10015314 | EMTEQ | EASA.IM.A.S.00184 | | FAA STC ST00221AT | B727-100, B727-200 | Installation of a Digital Flight Data Recorder Upgrade from 6 to 11 channels in accordance with Avias Engineering Report No 92AMJT-003 R-1, Revision B, dated May 4, 1993 or later | | | 1. This approval should not be extended to other aircraft of this model which other previously approved modifications are incorporated, unless it is determined by the installer that the interrelationship between this change and any other previo (...) | 07.10.2004 |
| 10015315 | EMTEQ | EASA.IM.A.S.00185 | | FAA STC ST02006AT | 727-100, 727-200 | Installation of an Allied Signal 18 Channel Flight Data Recorder (Upgrade) for Aircraft not containing a Flight Data Acquisition Unit in accordance with Avias Engineering Report No 92AMJT-040R-1, No Revision, dated March 15, 1999, Appendix A, master (...) | | | 1. This approval should not be extended to other aircraft of this model on which other previously approved modifications are incorporated, unless it is determined by the installer that the interrelationship between this change and any other previ (...) | 07.10.2004 |
| 10015645 | EMTEQ | EASA.IM.A.S.01609 | REV 1 | FAA STC ST02019AT | B767-300 | Installation of Passenger Cabin Interior Arrangement | | Avias Engineering Master Drawing List (MDL) No. 99GCS.019MDL Revision E, 17th February 2000. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 02.02.2009 |
| 10015316 | EMTEQ ENGINEERING | EASA.IM.A.S.00188 | | FAA STC ST02869AT | BOEING 767-300 | Installation of a passenger cabin interior arrangement in accordance with Master Drawing List ANST-048-MDL, Revision C dated 27/05/2004 or later EASA approved revision | | | 1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated.2. If transfer of this Certificate is requested under Part 21A.116 the STC wi (...) | 01.10.2004 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------|-------------------|----------------|----------------------|---|--|---------------------|--|--|---------------|
| 10015404 | EMTEQ ENGINEERING | EASA.IM.A.S.01115 | | FAA STC ST02780AT | BOEING 757-200 | Installation of Honeywell Mark V Enhanced Ground Proximity Warning System (EGPWS) | | -EMTEQ Engineering Master Drawing List (MDL) EMTQ-015B-MDL Revision D dated November 02, 2004 or later approved revision. -Flight Manual Supplement EMTQ-015B-FMS Initial Release or later approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 04.05.2005 |
| 10015405 | EMTEQ ENGINEERING | EASA.IM.A.S.01116 | | FAA STC ST02761AT | BOEING 757 BOEING 767 | Installation of Honeywell RESCU 406AF Emergency Locator Transmitter (ELT) | | EMTEQ Engineering Master Drawing List EMTQ-030-MDL Revision B dated January 27 2004 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 05.06.2005 |
| 10015407 | EMTEQ ENGINEERING | EASA.IM.A.S.01118 | | FAA STC ST02896AT | AIRBUS A319 SERIES AIRBUS A320 SERIES AIRBUS A321 SERIES | Installation of a Kannad Model 406-AF Emergency Locator Transmitter (ELT) | | Drawing No. 2-23982-001 Revision B dated July 7, 2004 or later FAA approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 26.10.2005 |
| 10015408 | EMTEQ ENGINEERING | EASA.IM.A.S.01122 | | FAA STC ST02896AT | FOKKER F.28 MARK 0100 | Installation of a Kannad Model 406-AF Emergency Locator Transmitter (ELT) in accordance with FAA STC ST02896AT dated 22/07/05 or later EASA-approved revision. | | Data package containing the following documents, and additional supporting data: 1. Master Drawing List for a KANNAD 406AF Emergency Locator Transmitter (ELT) installation on various fleet types, EMTEQ document 2-23982-001, Revision E.2. Structural S (...) | | 10.03.2006 |
| 10015566 | EMTEQ ENGINEERING | EASA.IM.A.S.01470 | | FAA STC SA03020AT | BAE JETSTREAM 3101 BAE JETSTREAM 3201 SERIES | Installation of Sandel Model ST3400 Terrain Awareness and Warning System (TAWS) | | EMTEQ Engineering Approved Model List (AML) Document No. 2-25975-002 Rev. IR dated January 14 2005 or later approved revision. EMTEQ Engineering AFM Supplement Document No. 2-25975-909 Rev. IR dated January 14 2005 or later approved revision. | | 10.01.2006 |
| 10015568 | EMTEQ ENGINEERING | EASA.IM.A.S.01475 | | FAA STC ST02793AT | BEECH 1900D | Installation of Honeywell Mk VI Enhanced Ground Proximity Warning System (EGPWS). | | EMTEQ Engineering Master Drawing List EMTQ-028-MDL, Revision B, dated 1st March 2004. Approved Flight Manual Supplement for a Honeywell Mk VI Enhanced Ground Proximity Warning System as installed in Beech Model 1900D, Revision IR, dated 24 February 2004. | 1. AFMS is required on-board for operation with Beechcraft 1900 D series. | 01.12.2005 |
| 10015569 | EMTEQ ENGINEERING | EASA.IM.A.S.01476 | | FAA STC ST03022AT | EMBRAER EMB-120 | Installation of Sandel Model ST3400 Terrain Awareness and Warning System | | FAA approved flight manual for STC ST03022AT, Doc. 2-25979-909EMTEQ JAA TGL-12 Compliance Report and E-mail December 12, 2005 | | 19.12.2005 |
| 10015570 | EMTEQ ENGINEERING | EASA.IM.A.S.01477 | | FAA STC SA02821AT | BEECH 1900D RAYTHEON 1900C | Installation of Sandel Model ST3400 Terrain Awareness and Warning System | | Emteq Engineering Master Drawing List EMTQ-036-MDL, Revision B, dated 19 March 2004. AFWS TAWS/TCAS No EMTEQ-036A-FMS, Rev IR, dated 26 March 2004 or AFMS TAWS/RMI No EMTEQ-036B-FMS, Rev IR, dated 26 March 2004. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. 2. AFMS is req (...) | 28.11.2005 |
| 10015571 | EMTEQ ENGINEERING | EASA.IM.A.S.01478 | | FAA STC ST03003AT | BOEING 767-200 BOEING 767-300 | Honeywell Mk V Enhanced Ground Proximity Warning System Installation | | EMTEQ Engineering Master Drawing List (MDL) 2-26425-001, Rev. A, dated December 05, 2004, or later approved revision. AFM Supplement for Boeing 767 series aircraft, EMTEQ Document No. 2-26425-908, Rev. IR, dated December 10, 2004, or later FAA approved revision. | | 01.12.2005 |
| 10015638 | EMTEQ ENGINEERING | EASA.IM.A.S.01595 | REV 1 | FAA STC ST02929AT | 1. BAE JETSTREAM 3101 2. BOEING 727-100 3. BOEING 737-100 4. BOEING 747-100 5. BOEING 757-200 SERIES 6. BOEING 767-200 7. BOEING 777-200 SERIES 8. DC-9-10 SERIES BOEING 727-200 BOEING 737-200 BOEING 737-300 BOEING 737-400 BOEING 737-500 BOEING 737-600 BOEING 737-700 BOEING 737-800 BOEING 747-200B BOEING 747-200C BOEING 747-200F BOEING 747-300 BOEING 747-400 SERIES BOEING 747-SP BOEING 767-300 DC-9-20 SERIES DC-9-30 SERIES DC-9-40 SERIES DC-9-50 SERIES | Installation of an ARTEX Model C406-1 ELT Emergency Locator Transmitter (ELT) in accordance with EMTEQ Master Drawing List(s) (MDL) as contained in EMTEQ Approved Model List (AML), Document No. 2-22566-002, Revision G, dated 09 June 2006, or later approved revision. | | | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. 2. This approv (...) | 12.09.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------|-------------------|----------------|----------------------|--|---|---|---|--|---------------|
| 10015828 | EMTEQ ENGINEERING | EASA.IM.A.S.01917 | | FAA STC ST03212AT | BOEING 737-300 SERIES BOEING 737-400 SERIES BOEING 737-500 SERIES | Installation of an Active Noise Reduction (ANR) Headset | | EMTEQ Engineering Master Drawing List (MDL) 2-31582-001, Rev. A, dated January 06, 2006 or later approved revision | | 23.05.2006 |
| 10015951 | EMTEQ ENGINEERING | EASA.IM.A.S.02093 | | FAA STC ST03269AT | BAE JETSTREAM 3101 BAE JETSTREAM 3201 SERIES | Installation of a Honeywell CAS 67A Traffic Alert & Collision Avoidance System II (TCAS II) & Dual Mode S Transponder | | EMTEQ Master Drawing List (MDL) 2-37239-001 Revision B dated May 12 2006 or later approved revision.AFM Supplement 2-37239-903 Revision IR dated 15 2006 or later approved revision. | | 01.08.2006 |
| 10016111 | EMTEQ ENGINEERING | EASA.IM.A.S.02301 | | FAA STC ST03345AT | BOEING 767-200 BOEING 767-300 | Installation of an Enhanced Surveillance System (EHS) to expand Transponder Mode-S functionality | | EMTEQ Master Drawing List (MDL) No. 2-38842-001, Rev. A, Dated October 03, 2006, or later approved revision.FAA Approved Airplane Flight Manual Supplement (AFMS) for an EHS Installation, as recorded in EMTEQ AML Document No. 2-38842-002, Rev. IR, or later approved revision | None | 26.02.2007 |
| 10016113 | EMTEQ ENGINEERING | EASA.IM.A.S.02303 | | FAA STC ST02297AT | AIRBUS A300 MODEL B2-203 AIRBUS A300 MODEL B4-103 AIRBUS A300 MODEL B4-203 | This installation added three additional parameters to be recorded by the existing Flight Data Recorder. | | Avias Engineering Master Drawing List No. 313AV30-MDL, Revision C, dated March 8, 2001, or later approved revision | | 21.06.2007 |
| 10016180 | EMTEQ ENGINEERING | EASA.IM.A.S.02409 | | FAA STC ST03394AT | BAE JETSTREAM 4101 | Installation of an Avionics Innovations DMP-100/200 Digital Media Player (DMP) for reproduction of Boarding Music and Cabin Briefings | | o EMTEQ Master Drawing List (MDL) No. 2-43923-001, Rev. Do Instructions for Continued Airworthiness; 2-43923-902 (Installation); 2-43923-910 (AMM); 2-43923-911 (IPC) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 30.01.2009 |
| 10016433 | EMTEQ ENGINEERING | EASA.IM.A.S.02773 | | FAA STC ST03462AT | BOEING 767-200 BOEING 767-300 BOEING 767-400 | Install the RAAS software in the Runway Awareness & Advisory System (RAAS) | | EMTEQ Engineering Master Drawing List (MDL) 2-41556-002, Rev IR, 12th June 2007.AFM Supplement, EMTEQ Document No. 2-41556-814, Rev IR, 07th November 2007. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.This (...) | 05.06.2008 |
| 10016434 | EMTEQ ENGINEERING | EASA.IM.A.S.02774 | | FAA STC ST03461AT | BOEING 757-200 BOEING 757-300 | Installation of the Honeywell Runway Awareness & Advisory System (RAAS). | | EMTEQ Engineering Master Drawing List (MDL) 2-41556-001, Rev IR, 12th June 2007.AFM Supplement, EMTEQ Document No. 2-41556-807, Rev IR, 27th November 2007. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 05.06.2008 |
| 10016658 | EMTEQ ENGINEERING | EASA.IM.A.S.03119 | | FAA STC ST03660AT | PILATUS PC-12 | LED Position Lights Installation/Removed Equipment:- Existing Left and Right Wing Tip, and Tail Position Lights/New Installed Equipment:- Red LED Position Light Assy, P/N 03-9000-1103- Green LED Position Light Assy, P/N 03-9000-2104- White LED Position Light Assy, P/N 03-9200-2001 | | - Certification Plan, Doc. No. 2-56640-912, Initial Release or latest approved Revision- Master Data List, Doc. No. 2-56640-001, Rev. A or latest approved Revision- ICA, Doc. No. 2-56640-905, Initial Release or latest approved Revision | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 21.01.2009 |
| 10033802 | EMTEQ ENGINEERING | | | FAA STC ST03282AT | B737-600, B737-700 B737-800, B737-900 | Installation of Dual Collins HFS-900 High Frequency (HF) communications system. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.CRI H-01 Issue 2: Enhanced Airworthiness Program for Aeroplane Systems- ICAon EWIS.Th (...) | Master Data List (MDL) Document No.2-36870-001 Revision D Dated January 10, 2011 Airplane Flight Manual Supplement (AFMS) Document No. 2-36870-849 dated June 19, 2006 or later revisions of the above listed documents approved by EASA in accordance with EAS (...) | None. | 09.02.2011 |
| 10035812 | EMTEQ ENGINEERING | | | FAA STC ST03906AT | CL-600-2B19 | Installation of EMTEQ LED Anti-collision/Position (Navigation) Light in the Bombardier Regional Jet Series 100/200/440 Aircraft | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | FAA STC No. ST03906AT dated 04 February 2011.EMTEQ Engineering Master Drawing List 2-83710-001 Revision B dated 28 December 2010 or later approved revision.Instructions for Continued Airworthiness (ICA), EMTEQ Engineering Document No. 2-83710-805 Rev (...) | None | 19.07.2011 |
| 10036838 | EMTEQ ENGINEERING | | REV. 1 | FAA STC ST03918AT | 777-200, 777-200LR 777-300, 777-300ER | FAA STC ST03918AT: Installation of the Honeywell MCS-7200 SATCOM System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.CS 25 Amtd 4CRI H-1 EWIS ICAThe certificated noise and/ or emissions levels of the orig (...) | FAA STC ST03918ATMDL 78144-001 Rev. A dtd. 13.May 2011-ICA 2-78144-804 Rev. IR dtd. 10.Feb 2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | Not Applicable | 17.10.2011 |
| 10037018 | EMTEQ ENGINEERING | | | FAA STC ST03253AT | 757-200, 757-200CB 757-200PF, 757-300 | Installation of the Honeywell MK II Aircraft Communication Addressing and Reporting System (ACARS). | FAR 25The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged an (...) | EMTEQ Engineering Master Drawing List 2-35716-001 Revision C, dated 16 November 2007.Flight Manual Supplement 2-35716-807 Revision IR, dated 20 April 2006, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Dec (...) | Compliance with EASA AMC 20-9 and AMC 20-10 has not been shown. | 25.10.2011 |
| 10037514 | EMTEQ ENGINEERING | | | FAA STC ST03606AT | B757-200/200CB/200PF/300 | Installation of the Iridium Communication System (ICS) | FAR-25The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and (...) | EMTEQ Engineering Master Drawing List 2-69511-001 Revision F01 dated 17 Nov 11 Flight Manual Supplements 2-69511-804 Revision A dated 15 Nov 11 2-69511-805 Revision A dated 15 Nov 11 or later revisions of the above listed documents approved by EASA in acc (...) | None | 13.12.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|--------------------|----------------|----------------------|--|--|---|--|---|---------------|
| 10038431 | EMTEQ ENGINEERING | | REV. 1 | FAA STC ST03345AT | B737-300/400/500 | Validation of FAA STC No. ST03345AT issued 27 October 2006. Installation of Elementary Surveillance System / Enhanced Surveillance System (ELS/EHS) in the Boeing 737-300/400/500 series aircraft. Installation of EHS capable Mode S transponders and ELS capable ATC/TCAS/Flight ID control panel. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. EASA Certification Review Item CR1 H-1 "Instruction for Continued Airworthiness on EWIS (...) | Master Drawing List Doc. 2-38842-001 for installation of ELS/EHS, Rev. L dated 11 June 2010. Approved Model List Doc. 2-38842-002 for installation of ELS/EHS, Rev. B dated 30 April 2008. Airplane Flight Manual Supplement Doc. No. 2-54717-901, Rev. IR, d (...) | None | 16.02.2012 |
| 10040185 | EMTEQ ENGINEERING | | | FAA STC ST03846AT | ATR 42-200/-300/-320 ATR 42-500 ATR 72-101/-102/-201/-202 ATR 72-211/-212/-212A | Installation of the IMS Flight Deck Ressources SkyTab 3201 Electronic Flight Bag (EFB) System. This STC is installing the SkyTab 3201 EFB mount arm/device and electrical supply, as well as a Satellite Weather Receiver together with antenna. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List, ref. 2-68788-001 Revision G, dated 26 March 2012. Instruction for Continued Airworthiness, ref. 2-68788-804 revision G, dated 26 March 2012. EASA Aircraft Flight Manual Supplement, ref. 2-68788-850, dated 06 April 2012 or later revision (...) | None | 15.06.2012 |
| 10042561 | EMTEQ ENGINEERING | EASA, IM.A.S.01015 | | | B737-200/-300/-400/-500 | Installation of Honeywell Ground Proximity Warning System | | -EMTEQ Engineering MDL EMTEQ-015A-MDL, Rev. J or later FAA approved revisions-FAA STC ST02716AT | 1. For Certification basis refer to FAA TCDS A16WE plus JAATGL No 12"TAWS"2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 19.01.2005 |
| 10042562 | EMTEQ ENGINEERING | EASA, IM.A.S.01016 | | | B737-300/400/500 | Installation of EMTEQ Engineering dual GPS Antenna Provisions | | - FAA STC ST02723AT- Avitas Engineering MDL Report No EMTEQ-012-MDL, Rev B or later FAA approved revisions | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 24.01.2005 |
| 10042953 | EMTEQ ENGINEERING | EASA, IM.A.S.01135 | | | B727 | Installation of a Honeywell Mark VII Enhanced Ground Proximity Warning System (EGPWS) according to FAA Supplemental Type Certificate No ST02995AT | | EMTEQ Engineering Master Drawing List Doc No 2-26236-001 Rev A dated November 20, 2004 or later FAA approved revision FAA Approved Flight Manual Supplement for Boeing 727 Series Aircraft, EMTEQ Doc No 2-26236-902 Rev IR dated November 20, 2004 or later FAA approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 30.05.2005 |
| 10043325 | EMTEQ ENGINEERING | EASA, IM.A.S.03181 | | FAA STC ST03678AT | BOEING 757-200 BOEING 757-300 | | Installation of the Spectralux Dlink + ACARS System in the Boeing 757-200/300 Series Aircraft | - Master Drawing List 2-59475-001 revision A, 22/01/2009. AFM Supplement, 2-59475-805, Revision A, 29/05/2009 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The (...) | 25.06.2009 |
| 10043542 | EMTEQ ENGINEERING | | REV. 1 | FAA STC 03980AT | 777-200, 777-200LR, 777-300, 777-300ER. | Installation of the Honeywell MCS-7147 SATCOM System. DESCRIPTION: Removes and Replaces the existing Honeywell MCS-6000 or MCS-7000 SATCOM System's Satellite Data Unit (SDU) and High Power Amplifier (HPA) with the MCS-7147 SATCOM System's SDU and HPA/High Speed Data Terminal (HSDT). | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: 25.23(a), 25.301(a), 25.561(c), 25.601, 25.603(a)(b)(c), 25.605(a), 25.60 (...) | - Master Drawing List 2-95373-001 Revision B, dated September 12, 2012- Instructions for Continued Airworthiness (ICA), EMTEQ Engineering Drawing No. 2-95373-804, Revision A, dated September 12, 2012 or later revisions of the above listed documents a (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 06.02.2013 |
| 10044694 | EMTEQ ENGINEERING | | REV. 2 | FAA STC ST03983AT | 777-200, 777-200LR, 777-300, 777-300ER, 777F. | Installation of the Cobham HGA-7001 High Gain SATCOM Antenna Sub-System. DESCRIPTION: Removes and Replaces the existing Canadian Marconi CMA 2102 High Gain SATCOM antenna with the Cobham HGA-7001 antenna | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Drawing List (MDL), EMTEQ document 2-106975-001, Revision B, dated September 19, 2012. Instructions for Continued Airworthiness (ICA), EMTEQ document 2-106975-803, Revision C, dated September 19, 2012 <(-,-,-) or later revisions of the above listed documents (...) | None | 03.05.2013 |
| 10044995 | EMTEQ ENGINEERING | | | FAA STC ST03916AT | A318-111 / -112 / -121 / -122 A319-111 / -112 / -113 / -114 A319-115 / -131 / -132 / -133 A320-111 / -211 / -212 / -214 A320-231 / -232 / -233 A321-111 / -112 / -131 / -211 A321-212 / -213 / -231 / -232 | Installation of the TEAC/Goodrich Digital Audio Reproducer and the TEAC/Goodrich Digital Video Reproducer for the Passenger Entertainment System (PES). | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC ST03916AT Issued May 6, 2011. Project Specific Validation Report 2-95307-907, Rev. IR, dated Mar. 4, 2013. MDL 2-95307-001, Rev. A, dated Apr. 28, 2011. Instruction for Continued Airworthiness 2-95307-805, Rev. A, dated Mar. 31, 2011. EWIS ICA 2-95307- (...) | None | 24.05.2013 |
| 10016117 | ENGINE TECHNOLOGIES, Inc. | EASA, IM.A.S.02309 | | FAA STC SA10588SC | CIRRUS SR22 | Validation of FAA STC SA10588SC "Installation of modified Teledyne Continental Motors IO-550-N engine equipped with turbo-normalizing system" | | Installation in accordance with Engine Technologies Master Drawing List ETI-G2-1, Rev. NC, dated 1. Nov 2006 or later FAA approved Revision | Installation of STC SE10589SC (EASA, IM.E.S.01013) is required. An oxygen system meeting requirements of CS23.1441 through 23.1449 must be installed in conjunction with this STC. The installer must determine, if this design is compatible with previously a (...) | 11.05.2007 |
| 10016699 | ENGINE TECHNOLOGIES, Inc. | EASA, IM.E.S.01013 | | SE10589SC | TELEDYNE IO-550-G | Installation of turbo-normalizing system in accordance with Engine Technologies, Inc. Master Drawing List Number ETI-G2-1, Rev. NC, dated November 1, 2006, or later FAA approved revision. | | Pilot's Operating Handbook and FAA Approved Airplane Flight Manual Supplement for Tornado Alley Turbo-normalizing System | 1. Approval of this change in type design applies to the above model engine only. This approval should not be extended to engines of this model on which other previously approved modifications are incorporated unless it is determined that the interre (...) | 02.04.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|-------------------------|---|--|--|--|--|---------------|
| 10046106 | ENGINE TECHNOLOGIES, Inc. | EASA.IM.A.S.02922 | | FAA STC SA5223NM | 1. 35-C33A, 36, A36, E33A 1. F33A, G33, S35, V35, V35A 1. V35B, A36TC, B36TC 2. G36 | Installation of modified Teledyne Continental engine equipped with Turbo-normalising System, optional Oxygen System and optional HartzellPHC-C3YF-1RF/F8068 propeller. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Installation in accordance with Tornado Alley Turbo Drawing List Number TAT-B-MDL Rev NC, dated September 29, 1999, or later FAA approved revision. -Operate in accordance with Flight Manual Supplements as defined in Limitations and Conditions section o (...) | See EASA approved Annex | 14.08.2013 |
| 10029711 | ERWIN AERO INTERNATIONAL LTD. | | REV. 1 | TCCA LSTC C-LSA09-321/D | DHC-6-300 | Avionics to be installed in acc. with PAA Design Group Modification Data Summary: R0910773, Issue A, approved 31 December 2009 with Transport Canada approved Flight Manual Supplements in acc. to TCCA STCC-LSA09-321/D | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplements: R0910775, Issue A, Collins DME-42 Distance Measurement Equipment-R0910776, Issue A, Collins ALT-55 Radio Altimeter-R0910777, Issue A, Collins ADF-60 Automatic Direction Finder-R0910778, Issue A, Collins TDR-90 Transponder- (...) | Limited to: S/N 840 only | 19.04.2010 |
| 10029999 | ESG ELEKTRONIKSYSTEM- | P-EASA.R.S.01419 | | | BK 117 C-2 | FLIR Mission Work Station PTA-TH installed on a FLIR StarSafire HD System and an Operator Console in accordance with ESG GmbH MDL 0005-MDL-006 based on ESG change no. 0005. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement No. 0005-FMS-025 Iss. 2 or later revisions of the above listed documents approved by EASA-Means of Compliance/ Certification Program 0005-MOCP-005 Iss. 2-Master Drawing List 0005-MDL-006, Iss. 3-Installation Instructions 0005-IO-024 | -ESG GmbH Minor Change 0003 and STC EASA.IM.R.S.01489 have to be installed as prerequisite for the installation of this modification.-This approval does not include the operation of laser pointer or laser illuminator devices. | 10.05.2010 |
| 10030259 | ESG ELEKTRONIKSYSTEM- | | | | MBB-BK 117 B-2 | Star SAFIRE and Mission Console BK117 installed on a FLIR Star Safire HD System on a bolster step on the RH side, mission console, equipment rack, operator and observer seats on adapter plates in the cabin, in accordance with ESG GmbH MDL0012-MDL-006 based on ESG change no. 0012 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement No. 0012-FMS-070 Iss. 3 or later revisions approved by EASA List of Compliance Documents 0012-LCD-003 Iss. 4 Means of Compliance Record 0012-MOCR-005 Iss. 3 Master Drawing List 0012-MDL-006, Iss. 3 Installation Instructions 0012- (...) | This approval does not include the operation of laser pointer or laser illuminator devices. | 07.06.2010 |
| 10042409 | ESG ELEKTRONIKSYSTEM- | EASA.R.S.01577 | | | EC135 P1, EC135 P2, EC135 P2+, EC135 T1, EC135 T2, EC135 T2+ | Trakka A800 Searchlight Installation Gimballed installation of a 800W searchlight and related electrical and mechanical provisions. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | EC135 Flight Manual Supplement "Trakka A800 Searchlight Installation" 0010-FMS-070, Rev. 0 or later revisions of the above listed documents approved by EASA- Instructions for Continued Airworthiness (AMM, WDM, IPC Supplements) 0010-ICA-081, Rev. 01- In (...) | -Activation and/or use of IR-Filter is not approved- Activation and/or use of Laser is not approved | 29.11.2012 |
| 10043040 | ESG ELEKTRONIKSYSTEM- | P-EASA.R.S.01011 | | | EC135 P1 (CDS) EC135 P1 (CPDS) EC135 P2 (CPDS) EC135 T1 (CDS) EC135 T1 (CPDS) EC135 T2 (CPDS) EC635 EC635 P2+ EC635 T2+ | FLIR Mission Work Station EC135 Optional installation of an operator station in the cabin for the operation of mission equipment. The installation is based on ESG Engineering Order 1RPA-ZL-EO_001 in accordance with ESG Master Drawing List 1RPA-MDL-036-01 Issue 1 or later approved revision. | EC135 Flight Manual Supplement: EC135 Flight Manual Supplement, 1RPA-FMS-025, Issue 1 Revision 2 or later approved revision. Installation Instructions and Instructions for Continued Airworthiness: Installation Instruction Order, Doc. No. 1RPA IIO-02 (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The (...) | 04.06.2009 | |
| 10016724 | EUROCOPTER CANADA Ltd. | EASA.IM.R.S.01022 | REV. 1 | TCCA STC SH96-39 | AS 350 BA/B2/B3 | Two place forward passenger seat (RH) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Refer to Master Drawing List MDL-96-004 issue 1 and associated installation procedure and maintenance instructions. | 1. Refer to Flight manual supplement attached to STC SH96-39 issue 5 approved 12 April 1996 or later approved revision. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph (...) | 19.12.2005 |
| 10016725 | EUROCOPTER CANADA Ltd. | EASA.IM.R.S.01023 | REV. 1 | TCCA STC SH 96-32 | AS350BA, AS350B2, AS350B3 | Left-side pilot configuration Note: Revision of EASA STC ref. EASA.IM.R.S.01023 being a validation of TCCA STC SH 96-32 at issue 7. This STC revision is validation of TCCASTC SH 96-32 at issue 11 and includes the left-side pilot configuration change ma (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Configuration and installation documentation: Eurocopter Canada Limited Master Drawing List No. MDL-96-003, Rev. U Maintenance documentation: Eurocopter Canada Limited Instructions for Continued Airworthiness No. ICA-ECL-168, Rev. 2 Operating documentation (...) | None | 13.12.2010 |
| 10016728 | EUROCOPTER CANADA Ltd. | EASA.IM.R.S.01031 | | TCCA STC SH 97-15 | AS 350 B/BA/B1/B2/B3/D | Installation of LH and/or RH Cargo pods (Squirrels Cheeks) Master Drawing List MDL-94-004 and associated maintenance instructions Flight manual supplement FMS-ECL-9, revision 2 or later TCCA approved revisions | | Master Drawing List MDL-97-004 and associated maintenance instructions Flight manual supplement FMS-ECL-9, revision 2 or later EASA approved revisions | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 12.10.2005 |
| 10016747 | EUROCOPTER CANADA Ltd. | EASA.IM.R.S.01073 | | TCCA STC SH 96-31 | AS 350 B/BA/BB/D/B1/B2/B3 AS 355 E/F/F1/F2 EUROCOPTER FRANCE AS 355F2 | Relocation of higher capacity battery to tail boom Master Drawing List MDL-04-003 and MDL-96-001 associated installation procedure and maintenance instructions. Flight Manual Supplement attached to STC SH96-31 issue 6, approved 25 March 1996 or later TCCA approved revision. | | Master Drawing List MDL-04-003 and MDL-96-001 associated installation procedure and maintenance instructions. Flight Manual Supplement attached to STC SH96-31 issue 6, approved 25 March 1996 or later EASA approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 12.10.2005 |
| 10016825 | EUROCOPTER CANADA Ltd. | EASA.IM.R.S.01195 | | TCCA STC SH00-1 ISSUE 3 | AS 350 B, AS 350 B1, AS 350 B2 AS 350 B3, AS 350 BA, AS 350 D | Installation of high visibility main and aft cabin doors refer to TCCA STC SH00-1 issue N°3, July 25, 2003 | | ECL installation procedure IP-ECL-42 rev 6 TCCA approved July 25, 2003 or later approved. ECL Maintenance Manual Supplement ECL-35 rev 0, TCCA approved August 12, 2002 or later TCCA approved | Refer to TCCA STC SH00-1 issue N°3, July 25, 2003 | 12.05.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|-------------------|----------------|--------------------------|--|--|--|--|---|---------------|
| 10016845 | EUROCOPTER CANADA Ltd. | EASA.IM.R.S.01221 | | TCCA STC SH05-18 | AS 350 B/B1/B2/B3/BA/D/D1 | Installation of Forward Short High-visibility Doors (350-202414). | | Installation must be in accordance with Eurocopter Canada Limited Installation Procedure n. IP-ECL-112, Rev. 0, TCCA approved Apr11, 2005 or later TCCA approved revisions. Configuration must be in accordance with Eurocopter Canada Limited Master Dra (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 05.07.2007 |
| 10016935 | EUROCOPTER CANADA Ltd. | EASA.IM.R.S.01321 | | TCCA STC SH97-34 ISSUE 3 | AS 350 B/B1/B2/B3/BA/D AS 355 E/F/F1/F2/N | Installation of Forward Opening Rear Cargo Doors. (Modification approved by TCCA under STC SH97-34 Issue 3). | | Definition and installation: ECL Master Drawing List No. MDL-06-002 Issue A Inspection and maintenance: ECL Instructions for Continued Airworthiness No. ICA-ECL-141 Rev. 0 | 1; This STC is approved only for the production configuration as defined in the approved design data referred to in the paragraph #Description#. Compatibility with other aircraft/engine configurations shall be determined by the installer. Installation (...) | 18.10.2007 |
| 10017000 | EUROCOPTER CANADA Ltd. | EASA.IM.R.S.01410 | | TCCA STC SH03-33 | EC 130 B4 | Cargo Pods Installation | | Definition and installation: ECL Master Drawing List MDL-03-013 Issue CECL Installation Procedure IP-ECL-109 Revision 1 Inspection and maintenance: ECL Instructions for Continued Airworthiness ICA-ECL-123 Revision 2 Compliance: Compliance Program CP-0253 Issue D | | 12.06.2008 |
| 10038250 | EUROCOPTER CANADA Ltd. | | | TCCA STC SH94-29 | 1- AS350B, BA, B1, B2, B3, D 2- AS355E, F, F1, F2, N, NP | Title: Left-Hand Litter Kit Description: Installation of Left-Hand Litter Kit P/N 350-200034 or P/N350-200194 in accordance with Eurocopter Canada Master Drawing List No. MDL-99-001. The Left-Hand Litter Kit provides med-evac capabilities for non-EMS (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FMS LEFT-HAND LITTER INSTALLATION (P/N's 350-200034 and 350-200194) ECL-107 Rev. 5 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision). Instructions for (...) | Configuration must be in accordance with Eurocopter Canada Master Drawing List No. MDL-99-001, Issue F, or later approved revisions. Approved configurations are listed in the attached table. | 31.01.2012 |
| 10041637 | EUROCOPTER CANADA Ltd. | | | TCCA STC SH12-10 | EC130B4 | Door handle installation (P/N 120-200014) This STC is the validation of TCCA STC SH12-10 (issue date - February 10, 2012) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Definition and installation: ECL Master Drawing List - doc. ref. MDL-10-015 Issue B Installation Procedure - RH Door Handle Installation - EC130B4 # doc.ref. IP-ECL-127 Rev. 2 Inspection and Maintenance: Instructions for Continued Airworthiness - RH Doo (...) | None | 03.10.2012 |
| 10041736 | EUROCOPTER CANADA Ltd. | | | TCCA STC SH11-35 | EC130B4 | Glare Shield Modification P/N 130-85114 This STC is the validation of TCCA STC SH11-35 (issue No. 1 dtd September 20, 2011) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Operation: Eurocopter Canada Limited Flight Manual Supplement - Adjustable Glare Shield Mod EC130 - doc. ref. FMS-ECL-148 Rev. 0 Definition and installation: Eurocopter Canada Limited Master Drawing List - Adjustable Glare Shield Mod EC130B4 - doc. ref. MD (...) | None | 11.10.2012 |
| 10043307 | EUROCOPTER CANADA Ltd. | | | TCCA STC SH12-9 | EC120B | Door Handle installation (P/N 120-200014) Validation of TCCA STC SH12-9 Issue 1 dated 09/02/2012 to correct handling problems induced by the serial design of the right-hand cabin door handle, Eurocopter Canada Ltd has designed a new door handle. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Definition and Installation in accordance with: Eurocopter Canada Limited Master Drawing List MDL-10-014, Issue B- Eurocopter Canada Limited Installation Procedure IP-ECL-126, Revision 1 Maintenance in accordance with: Eurocopter Canada Limited Instr (...) | None | 18.01.2013 |
| 10043426 | EUROCOPTER CANADA Ltd. | | REV. 1 | TCCA STC SH96-126 | AS 350B, AS 350D, AS 350B1, AS 350B2, AS 350BA, AS 350B3, EC 130 B4, AS 355 E, AS 355 F, AS 355 F1, AS 355 F2, AS 355 N, AS 355 NP | Blade Tie-Down Provisions as defined in Eurocopter Canada MDL-96-012 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Eurocopter Canada MDL-96-012 Rev. F or later TCCA approved revisions. Eurocopter Canada ICA-ECL-237 Rev.1 or later TCCA accepted revisions. | This modification is compatible only with main rotor blades equipped with tracking fingers p/n 350A11-1248-20 or 355A11-2013-20. Prior to installation of this design change it must be determined that the interrelationship between this design change and (...) | 07.02.2013 |
| 10045652 | EUROCOPTER CANADA Ltd. | | REV. 1 | TCCA SH10-20 | EC130B4 | Operation with the following cabin door configuration: LH pilot door closed, LH sliding door open and locked, RH doors/panels removed. This STC is the validation of TCCA STC SH10-20 (Issue date April 22, 2010) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Eurocopter Canada Limited Flight Manual Supplement FMS-ECL-137 rev. 0 or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement. Eurocopter Canada Limit (...) | This approval of operation with cabin doors / panel removed does not constitute approval of installation of those doors / panel being removed. Installation of all doors / panels being removed in order to obtain the configuration covered by this approval (...) | 12.07.2013 |
| 10033870 | EUROCOPTER DEUTSCHLAND GmbH | P-EASA.R.S.01568 | | | EC120B | Installation on an EC120B helicopter of a ETSO-certified (ETSO C113) Moving Map system (MT Vision Air EPIII) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Flight Manual Supplement, SK-MIC-110-SUP-P-245 dated 1/10/2010 or later revisions of the above listed documents approved by EASA - Declaration of Compliance, SK-MAC-106-DC-P-187 dated 4/10/2010 - Compliance Report, SK-MIC-110-CRP-P-246A dated 30/9/2010 | None | 15.02.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|----------------|----------------|---------------------------|---|--|--|---|--|---------------|
| 10039633 | EUROCOPTER DEUTSCHLAND GmbH | | | LBA STC EMZ NR. 0654/3058 | BO105 LS A-3 | Title: Superlifter Conversion Kit 105-8003331 Description: Original STC: The model BO105 LS A-3 is changed in accordance with drawing No. 105-800331. The changed model is called "Superlifter". Revision 1: The Airworthiness Limitations related to EMZ Nr. 0654/3058 has been changed and approved. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Original STC: Flight Manual BO 105 LS A-3, DOT-approved dated 25th February 1991 or later DO1 approved versions until 29th June 2009 FMA 12-1 "Appendix for BO 105-LS A-3 Superlifter Operator". LBA-approved dated 6th October 1995 or later LBA approved ve (...) | The implementation of the modified cargo hook 105-803101 is only approved after conversion of the helicopter in accordance with drawing No. 105-800331. This STC is valid in conjunction with the EASA TCDS R.011 in the applicable version. This Supplemental (...) | 24.05.2012 |
| 10041899 | EUROCOPTER DEUTSCHLAND GmbH | | | | EC 120 B | Installation of GTN 150 Unit on EC 120B | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - SK-MIC-136-SUP-P-623 Flight Manual Supplement, Revision B, dated 08-2012 or later revisions of the above listed documents approved by EASA- P347-001 (00000-00 GTN Installation (Top Drawing)- SK-MIC-136-EO-P-616 Engineering Order- SK-MIC-136-CM-P-620 Customization Manual | none | 23.10.2012 |
| 10017184 | EUROCOPTER ESPAÑA (ECE) | EASA.R.S.01317 | | | EC 135 P2 (CPDS), EC 135 P2+ EC 135 T2 (CPDS), EC 135 T2+ | Daylight Camera Installation (Ultramedia III) (Structural and Electrical Provisions): Installation of a camera Ultramedia II/III/HD in accordance with ECE Master Document List DL L853H0001E01, Rev. B. | | Flight Manual Supplement: "Supplement for FSI UM II/III/HD Gyro Stabilised Camera", FMA 11.100. Installation Instructions: Master Document List DL L853H0001E01, Rev. B. Instructions for Continuous Airworthiness "Flir Manual FSI UM II/III/HD", Rev. A. | | 14.04.2008 |
| 10017202 | EUROCOPTER ESPAÑA (ECE) | EASA.R.S.01351 | REV. 1 | | EC-135-P2(CPDS), EC-135-P2+ EC-135-T2 (CPDS), EC-135-T2+ | Daylight Camera Installation (MX-15i Wescam) (Structural & Electrical Provisions) and Operator Work Station. Installation of a Camera MX15i Wescam and related Operator Work Station in accordance with ECE Master Document Lists DL L853H0002E01 Rev. B and DL L854H0001E01 Rev. B. | | Flight Manual Supplement: Supplement for Forward Looking Infrared Wescam System LX-15i FMA 11.101 Installation Instructions: Master Document Lists DL L853H0002E01 Rev. B and DL L854H0001E01 Rev. B. Instructions for continuous Airworthiness "Flir Manual Wescam MX-15i" Rev. A. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 14.04.2008 |
| 10017128 | EUROCOPTER UK Ltd. | EASA.R.S.01216 | | | EUROCOPTER FRANCE AS 355N | Integration of minor and major changes to Storm Light, 406MHz ELT, STBY Altimeter, Exhaust heat shield, Autopilot, ADI, COM/NAV, Transponder, Audiopanel, GNS 530, Compass System and Instrument Panel | | -MCH/A/355/691 | 1. Limited to AS355N, S/N 5708 OE-XDM2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 13.10.2005 |
| 10017136 | EUROCOPTER UK Ltd. | EASA.R.S.01230 | REV. 1 | | EC135-P1+, EC135-P2+ EC135-T1, EC135-T2, EC135-T2+ | Change to existing EASA STC EASA.R.S.01230 - NVG Compatible Lighting System - to extend the approved Night Vision Goggles types to allow single pilot operations and to include the latest EC135 types. | As per EC135 TCDS The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Modification Definition Design Report DR/135900-001 Issue C dated 2 September 2009 Flight Manual Supplements- ECUK RFMS No. 28 Issue 2 dated 18 November 2009 For dual pilot operations using Northrop Grumman M949 NVG on EC135 T1, T2 and T2+- ECUK RFMS No. 29 (...) | None | 20.11.2009 |
| 10017177 | EUROCOPTER UK Ltd. | EASA.R.S.01295 | | | EUROCOPTER DEUTSCHLAND MBB-BK | Variable External Payload Configuration | | MHL BK 117 C-2 RFM Supplement No. 2 (Wescam MX-15) MHL BK 117 C-2 RFM Supplement No. 3 (NAT Skysubout) | Refer to above RFM Supplements | 16.03.2007 |
| 10017199 | EUROCOPTER UK Ltd. | EASA.R.S.01348 | | | EUROCOPTER DEUTSCHLAND EC-135- | Installation of Honeywell Avionics MKXXII EGPWS | | Design Report MCH/A/135/986 Rev BAFM Supplement No.34 to EC135T2 AFM Rev B | As defined in the AFM Supplement No. 34 | 03.04.2007 |
| 10017219 | EUROCOPTER UK Ltd. | EASA.R.S.01376 | | | EUROCOPTER FRANCE AS 365N3 | Re-Instatement of approval for Category A Operations reference FAR 29 Amendment 16 | | EASA STC 2005-988. UKCAA AAN28812. McAlpine Helicopters Change Reference MCH/A/365/726 McAlpine Helicopters Project 0223 AS365N3 Serial Number 6846. McAlpine Helicopters RFM Supplement No.6 Honeywell CDS-R Cockpit Display System | | 13.04.2007 |
| 10017253 | EUROCOPTER UK Ltd. | EASA.R.S.01456 | | | EUROCOPTER FRANCE AS 365N3 | Installation of NAT 611-014 Voice Warning Generator (AVAD). | | - Eurocopter UK. Change Number: MHL/365/1110. Installation of NAT 611-014 Voice Generator (AVAD)- Eurocopter UK. Design Report Number: DR/365110-001 issue A, dated February 14th, 2008. - Eurocopter UK FMS AS365N3 Number 30, dated February 14th (...) | | 22.04.2008 |
| 10017256 | EUROCOPTER UK Ltd. | EASA.R.S.01463 | REV. 1 | | EUROCOPTER DEUTSCHLAND EC-135- | Installation of FreeFlight 2101 I/O Approach Plus GPS with navigation information interfaced to the aircraft FCDS and the aircraft AFCS for coupled autopilot operations, law Drawing No. A3/1351114-100. | | - Flight Manual Supplement FMS "EC UK EC135T2+ FMS No. 36" Issue 1, 2 Installation Instructions, Instructions for Continued Airworthiness: Maintenance Requirements Document MR/1351114-001 Issue A, dated 18 March 2008 or later revisions. - Waste (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 11.12.2008 |
| 10017260 | EUROCOPTER UK Ltd. | EASA.R.S.01469 | | | EUROCOPTER DEUTSCHLAND EC-135 | Installation of honeywell EGPWS Mk. XXII. Installation of Honeywell EGPWS XXII law drwg. no. A3/1351134-100 with aural and display based on ECUK modification MHL/135/1134. | | Flight Manual Supplement FMS Nr. 40 "Honeywell Mk XXII EGPWS" issue 1. Maintenance Requirements Document MR/1351134-001, issue C, dated 14-07-2008 or later approved revision. Master Drawing no. A3-1351134-100, latest revision. | ECUK modification MCH/A/135/1075 must be installed as prerequisite. | 01.09.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------|------------------|----------------|----------------------|--------------------------------|--|--|--|---|---------------|
| 10017264 | EUROCOPTER UK Ltd. | EASA.R.S.01475 | | | EUROCOPTER DEUTSCHLAND EC-135- | Installation of STAR SAFIRE HD FLIR (based on EC UK modification MHL/135/1116).. | | # Flight Manual Supplemento EC UK EC135T2+ EC135P2+ FMS No. 44 Issue 10 If in addition the Retractable Digital Microwave Downlink Antenna, i.a.w. Eurocopter UK Ltd Change No. MCH/A/135/791 is installed EC UK EC135T2+ EC135P2+ FMS No. 42 Iss (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 14.11.2008 |
| 10017269 | EUROCOPTER UK Ltd. | EASA.R.S.01489 | | | EUROCOPTER FRANCE AS 355NP | Installation of a Leo II FLIR System | | Flight Manual Supplement No 18 # Leo II FLIR System # issue 1 or later EASA approved revision; Maintenance Manual Supplement MMS/3551127-001 issue A dated 5 June 2008 or later EASA accepted revision; Top Level Installation Drawing A3/3551127-100 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 13.01.2009 |
| 10017270 | EUROCOPTER UK Ltd. | EASA.R.S.01490 | | | EUROCOPTER FRANCE AS 355NP | Installation of SFIM 85T31 3 Axis Autopilot System | | Flight Manual Supplement No 29 # SFIM 85T31 (3 Axis) Autopilot VFR # Issue 1 or later EASA approved revision; Maintenance Requirements Document MR/3651130-001 issue 1 or later EASA approved revision; A3/3551130-100 # GA of 3 Axis Autopilot System # Issue A | 0 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 14.01.2009 |
| 10017271 | EUROCOPTER UK Ltd. | EASA.R.S.01491 | | | EUROCOPTER FRANCE AS 355NP | Installation of an alternate Pilot System | | Eurocopter UK Ltd. AS355NP Flight Manual Supplement No.28 dated 3 November 2008 or later EASA approved revision. Eurocopter UK Ltd. AS355NP drawing A3/3551133-100-1 # GA of Alternate Pilot System # Issue A Eurocopter UK Ltd. AS355NP report MR/3651133-0 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 11.12.2008 |
| 10017281 | EUROCOPTER UK Ltd. | EASA.R.S.01529 | | | EUROCOPTER DEUTSCHLAND EC-135- | FSI STAR SAFIRE FLIR SKID MOUNTED Optional installation of an external FLIR camera FSI Star Safire on the right hand multifunction carrier. The installation is based on EC UK change MHL/135/1155 and in accordance with drawing number A4/1351155-100-1 Issue A or later approved revision. | | 2 Flight Manual Supplemento EC UK EC135T2+ EC135P2+ FMS No. 50 Issue 1; 2 Installation Instructions, Instructions for Continued Airworthiness o Maintenance Requirements Document MR/1351155-001 Iss. A, or later approved revision o Master D (...) | 1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2) Th (...) | 09.06.2009 |
| 10017284 | EUROCOPTER UK Ltd. | EASA.R.S.01532 | REV. 1 | | EC 135 T2+, EC 135 P2+ | Installation of Spectrolab Nightsun XP Skid Mounted System. Equipment inspection and recertification to close the requirement for AD No.2010-0183R1. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Maintenance Requirements Document MR/1351151-001, Issue B, dated 03/11/2010 Design Report DR/1351151-001, Issue B, dated 11.11.2010 | None | 15.11.2010 |
| 10017288 | EUROCOPTER UK Ltd. | EASA.R.S.01537 | | | EUROCOPTER DEUTSCHLAND BK117 C | - Installation of a MX 15 FLIR. Skid Mounted (Meeker Mount)- Installation of a MX 15 FLIR in combination with the installation of EASA.IM.R.S.01489. The installation is based on EC UK MHL/145/1157 i.a.w. Drw. No. A2/1451157-100. | | - Rotorcraft Flight Manual Supplement: # Supplement No. 05 "WESCAM MX15 FLIR Surveillance System", Issue No. 1 or later approved revision- Instructions for Continued Airworthiness and Installation Instructions: # Maintenance Requirements D (...) | 1. The EC UK change MHL/145/1001 must be installed prior to the installation of MHL/145/1157.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previous (...) | 31.03.2009 |
| 10017294 | EUROCOPTER UK Ltd. | EASA.R.S.01578 | | | EUROCOPTER DEUTSCHLAND EC-135- | Bucher 16g Stretcher Installation Installation of Bucher Leichtbau 16g Stretchers and Associated Equipment in the Rear Cabin. The installation is based on EC UK Major Change MCH/A/98/40H, in accordance with Master Drawing No. A0/98490-100 Issue B. This STC supersedes CAA UK AAN 26855 at Issue 2. | | Flight Manual Supplements:- When installed in EC135 T1 (CDS), T1 (CPDS); McAlpine Helicopters Ltd. Supplement No. 5, Bucher 16g Stretcher Installation, Issue 1 or later approved revision- When installed in EC135 T2 (CPDS); McAlpine Helicopters (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 15.06.2009 |
| 10027764 | EUROCOPTER UK Ltd. | P-EASA.R.S.01588 | | | EC135 T2+, EC135 P2+ | Title: Installation of an External Camera and related ENG equipment Description: Installation of an Axsys Technologies Cineflex V14 Camera based on EC UK change MHL/135/1200 as defined in Drw. No. A3/1351200-100-1 Issue A and installation of the relate (...) | JAR27 First Issue as per TCDS EASA.R.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement: EC135 FMS No. 53 Rev. 0 or later revisions approved by EASA/ICA; Maintenance Requirement Document MR/1351200-001 Issue A; Maintenance Requirement Document MR/1351198-001 Issue C | None | 30.10.2009 |
| 10028856 | EUROCOPTER UK Ltd. | | | | BK117 C-2 | Cabin Configuration Definition of permissible configurations and combinations of individually approved changes based on EC UK Major Change MHL/145/1229 Iss. A into the BK117 C-2. | The Certification Basis for the original product remains applicable to this certificate/ approval. | BK117 C-2 EC UK Flight Manual Supplement No. 14 "Cabin Configuration", Iss. 1 or later revisions of the above listed documents approved by EASA-Design Report DR/1451229-001 Iss. A | None | 16.02.2010 |
| 10028925 | EUROCOPTER UK Ltd. | P-EASA.R.S.01590 | | | BK117 C-2 | Bucher 16G stretcher | FAR 29 The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual, EC UK BK117-C2 (EC145) Supplement No.11 or later revisions of the above listed documents approved by EASA-Build standard/top drwg; A3/1451194-01-1-ICA; Maintenance requirements MR/145/1194-001 | None | 19.02.2010 |
| 10029165 | EUROCOPTER UK Ltd. | P-EASA.R.S.01591 | | | MBB-BK117-C2 | Martin Baker Swivelling Passenger Seats | EASA TCDS R.010 The Certification Basis for the original product remains applicable to this certificate/ approval | ECUK RFMS Number 9, Issue 1 or later revisions of the above listed documents approved by EASA-Eurocopter UK Design Report DR/145/1196 Iss A. | None | 09.03.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------|----------------|----------------|----------------------|---|--|---|---|--|---------------|
| 10029931 | EUROCOPTER UK Ltd. | | REV. 1 | | MBB-BK117 C-2 | Cabin Configuration - Installation of Police or Emergency Medical System/Role Equipment | The Certification Basis for the original product remains applicable to this certificate/ approval | Flight Manual Supplement No. 16, Issue No. 1-Design Report DR/1451230-001 Issue A dated 15 May 2010 or later revisions of the above listed documents approved by EASA | None | 11.06.2010 |
| 10030287 | EUROCOPTER UK Ltd. | | | | MBB-BK 117 C-2 | Wescam HDI FLIR camera system. Installation of Wescam HDI FLIR camera system with build standards for alternate installation of MX-15 Analogue system in accordance with EC UK change no. MHL/145/1211, Master Drawings A3/1451211-100-1&2. The installation (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. | Eurocopter UK BK117 C-2 (EC145) Supplement No. 18, Issue 1 or later revisions of the above listed documents approved by EASA - Instructions for Continued Airworthiness - ICAW/1451211-001, Iss. A - Design Report DR/1451211-001, Iss. A | 1. Following items must be installed on the aircraft prior to incorporation of this change: ACS/MEEKER Utility step on the right hand side, p/n AFM/SM117A-11 and related Arm p/n AF-SM117-C10 installed i.a.w. STC EASA.I.M.R.S.014892. This approval does not (...) | 09.06.2010 |
| 10030343 | EUROCOPTER UK Ltd. | | | | BK117 C-2 (EC145) | Dual Martin Baker Swivelling Passenger Seats | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ECUK RFMS Number 19 Issue 1 or later revisions approved by EASA-DR/145/1196-002 Issue A | As identified in RFMS 19 | 11.06.2010 |
| 10031371 | EUROCOPTER UK Ltd. | | | | AS350B3 | Coupling of the existing Sagem/SFIM Autopilot installation with the Garmin GNS530 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Drawing List A4/3501234-001, issue A, dated 10.08.2010 - Instructions for Continued Airworthiness ICAW/3501234-001, issue B, dated 29.07.2010 - Supplement No. 17 to the Approved Flight Manual AS 350B3 Helicopter, issue 1, dated 3.08.2010 | none | 12.08.2010 |
| 10031633 | EUROCOPTER UK Ltd. | | | | MBB-BK 117C2 | Skyscout System Installation of a Skyscout System. With internal cabin control and external loudspeaker array, installed on aircraft RHS and configured with low skid landing gear | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement No 17, dated 24th August 2010 or later revisions of the above listed documents approved by EASA - Design Report DR/1451232-001 Issue A, dated 4th March 2010 | none | 01.09.2010 |
| 10032268 | EUROCOPTER UK Ltd. | | REV. 1 | | EC135 T2+ | NVIS Certification for EC135 T2+ REV. 1 is implementing administrative changes | as identified in EASA TCDS R.009FAA AC 27 at Change 2 MG16 Certification Procedures for Rotorcraft Night Vision Imaging Systems Lighting Equipment and RTCA documents DO275 and DO268/JAA Administrative and Guidance Material TGL No.34 Night Vision Imaging System Operations, dated 1 June 2003 | Eurocopter UK Rotorcraft Flight Manual Supplement No 56 Eurocopter UK Instructions for Continued Airworthiness ICAW/1351242 Issue A or later revisions of the above listed documents approved by EASA - Eurocopter UK Design Report DR/1351242 Issue A | None | 11.02.2011 |
| 10033580 | EUROCOPTER UK Ltd. | | | | EC 135 T1 (CDS) EC 135 T1 (CPDS) EC 135 T2 (CPDS) EC 135 T2+ | Installation of Spectrolab Nightsun XP on EC135 equipped with ECUK Equipment pod. Equipment inspection and re-certification to close the requirement for AD No. 2010-0183R1. Note: Identical to STC with ref. 10032575, re-issued for administrative purposes. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Maintenance Requirements Document MR/135787-001, Issue C, dated 05.11.2010 Design Report DR/135787-001, Issue E, dated 02.11.2010 | None | 16.11.2010 |
| 10034086 | EUROCOPTER UK Ltd. | | REV. 1 | | EC 130 B4 | Installation of Sagem Cockpit Display System MHL/130/1245 and revision to Eurocopter UK EC130B4 Supplement No. 1 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Eurocopter UK EC130B4 RFM Supplement No. 1 Issue 2 dtd 04.03.2011 - GPS Garmin GNS 530W Eurocopter UK EC130B4 RFM Supplement No. 3 Issue 1 dtd 04.03.2011 - Sagem Integrated Cockpit Display System (ICDS) or later revisions of the above listed documents a (...) | Prerequisite changes: MHL/130/1246, ..., TWX670 Weather detection system MHL/130/1070, ..., KRA405B Radio Altimeter MHL/130/1069, ..., Avidyne TAS620 Traffic Advisory System MHL/130/952, ..., Replacement of GNS430 with GNS530W/STC EASA.I.M.R.S.01061, Installation of Sagem Autopilot EC130 | 04.03.2011 |
| 10039454 | EUROCOPTER UK Ltd. | | | | EC135 P1 | Installation of Kelvin 275 Camera per Drawing List DL/1352044-001, Issue A | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement EC 135 P1 Supplement No. 58, Issue 1 or later revisions of the above listed documents approved by EASA - Instructions for Continued Airworthiness ICAW/1352044-001 Issue A - Drawing List DL/1352044-001, Issue A | -Day/Night VFR operation only - Operation of the Kelvin 275 camera approved under this STC (EC UK modification EC UK/135/2044) has to be done, in conjunction with EC UK modifications EC UK/135/2045 and EC UK/135/2043 | 04.05.2012 |
| 10039704 | EUROCOPTER UK Ltd. | | | | EC 135 P1 | Cabin Configuration Installation of equipment rack and pedestal mounted LCD monitor in the forward cabin, as per ECD design change EC UK/135/2043 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | EC UK RFM Supplement N°59, dated 10 May 2012 or later revisions of the above listed documents approved by EASA - Design Report N° DR/1352043-001, Issue B, dated 09 May 2012 - Instructions for Continued Airworthiness ICAW/1352043-001 | None | 18.05.2012 |
| 10043089 | EUROCOPTER UK Ltd. | EASA.R.S.01592 | | UK CAA AAN 27085 | EC135P2+, EC135T1, EC135T2 EC135T2+ | Installation Martin Baker Swivelling passenger Seats Note: This STC was issued to extend UK CAA AAN 27085 issued to McAlpine Helicopters Ltd. to the models EC135 P2+ and EC135 T2+. | | Eurocopter UK Supplement 6 issue 1 part of Eurocopter UK change referenced MCH/A/99/496/HUK CAA AAN No 27085 Issue 3 provided original approval. EASA STC to be issued to include addition of T2+ and P2+ variants. | none | 29.06.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|----------------------|--|---|--|---|---|---------------|
| 10043090 | EUROCOPTER UK Ltd. | EASA.R.S.01579 | | UK CAA AAN 28667 | EC135 P2(CPDS) - P2+, T2+ | NAT Skyshout Speaker System InstallationOptional installation and integration of the NAT PA700 Skyshout system. The modification is based on EC UK change MCH/A/135720 in accordance with Master Drawing A2/135720-100 at Issue H or later approved revision (...) | | Flight Manual Supplements:- When installed in in EC135 P2(CPDS): McAlpine Helicopters Ltd Supplement No. 1, NAT PA700 Skyshout External Loudspeakers, Issue 1 or later approved revision:- When installed in EC135 T2+/P2+ Eurocopter UK Ltd. Supplement N (...) | none | 15.06.2009 |
| 10043181 | EUROCOPTER UK Ltd. | | | | EC 135 T2+ | ECUK/135/2047 Installation of RTE Aircraft: Installation of LH and RH Meeker steps, EASA STC EASA.IM.R.S.01490 refers. AnSWE Systems Camera system is installed with Turret attached to the front of the Left Hand Meeker Step via Meeker A11 Utility Mount, (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Design Report N° DR/1352047-001, Issue A, dated 7 December 2012.- Instructions for Continued Airworthiness ICAM/1352047-001, Issue A, dated 14th December 2012.- Supplement N° 60 to the approved EC 135 T2+ RFM: "Installation RTE Aircraft", Issue 1, dated 18th December 2012. | For Day/Night VFR only. | 09.01.2013 |
| 10044632 | EUROCOPTER UK Ltd. | | | | EC135 P2 (CPDS), EC135 P2+, EC135 T2 (CPDS), EC135 T2+ | Installation of Ultraforce 350 Camera | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | RFM Supplement N°62 # Ultraforce 350 Camera Turret on Low Landing Gear, Issue 1 or later revisions of the above listed documents approved by EASA.- Design Report N° DR/1352065-001, Issue B, dated 11 April 2013.- Instructions for Continued Airworthine (...) | - Installation is limited to helicopters with low landing gear configuration.- The following Eurocopter Deutschland changes have to be embodied: E-1343, E-1482, E-1720, E-2144 | 25.04.2013 |
| 10043114 | EXECUTIVE JET GROUP LTD | EASA.A.S.02711 | | | 900 | Approval of Cabin Configuration & Changes for the Issue of an EASA C of A on Aircraft Reg G-HMEV | | | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall (...) | 26.06.2006 |
| 10040937 | F&M ENTERPRISES INC. | | | FAA STC SE7559SW | 1. C-75 SERIES, C-85 SERIES 2. C090 SERIES, O-200 SERIES | Install adapter to permit use of a spin-on oil filter in accordance with F. & M. Enterprises' Installation Instructions for Model TAF-Long Oil Filter, dated October 26, 1998, or later FAA approved revision. EASA Validation of FAA STC SE7559SW | The Certification Basis for the original product remains applicable to this certificate/ approval. | Installation Instructions for Model TAF-Long Oil Filter, dated October 26, 1998/ FM2007-01 Instructions for Continued Airworthiness, Rev. B, dated May 09, 2011 for later revisions of the above listed documents approved by EASA in accordance with EASA ED De (...) | Engines modified in accordance with this STC will not differ in performance or operating limitations for unmodified engines. Only the following filters are eligible for this installation: Champion No. CH48108A, C No. OF-53A | 06.08.2012 |
| 10040938 | F&M ENTERPRISES INC. | | | FAA STC SE09356SC | 1. O-470 SERIES 2. IO-470 SERIES 3. IO-520 SERIES | Install F. & M. Enterprises' Model C6LC, C6LC-S, or C6LC-L oil filter adapter for a spin on oil filter in accordance with F. & M. Enterprises' Installation Instructions dated October 26, 1998, or later FAA approved revision. EASA Validation of FAA STC SE09356SC | The Certification Basis for the original product remains applicable to this certificate/ approval. | C6LC<(->)> C6LC-S, C6LC-L, Installation Instructions dated Oct 26, 1998/ FM2007-01 Instructions for Continued Airworthiness, Rev. B, dated May 09, 2011 for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 20 (...) | None | 06.08.2012 |
| 10040939 | F&M ENTERPRISES INC. | | | FAA STC SE08409SW | 1. C-125 SERIES 2. C-145 SERIES, O-300 SERIES 3. GO-300 SERIES 4. IO-360 SERIES 5. TSIO-360 SERIES | Installation of F. & M. Enterprises Model CO-300 or C6SC spin-on oil filter adapter, in accordance with F. & M. Enterprises installation instructions for Model CO-300 dated October 26<(->)> 1998, or Model C6SC dated October 5, 2005, or later FAA approv (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. | Installation Instructions for Model CO-300 dated October 26<(->)> 1998/ Installation Instructions for Model C6SC dated October 5, 2005/ FM2007-01 Instructions for Continued Airworthiness, Rev. B, dated May 09, 2011 for later revisions of the above listed (...) | Engines modified in accordance with this STC will not differ in performance or operating limitations for unmodified engines. | 06.08.2012 |
| 10015468 | F. ATLEE DODGE AIRCRAFT SERVICES | EASA.IM.A.S.01221 | | FAA STC SA02115AK | PIPER PA-12 PIPER PA-12S PIPER PA-14 PIPER PA-18-150 PIPER PA-18A-150 PIPER PA-18AS-150 PIPER PA-18S-150 | Fabrication and installation of F. Atlee Dodge 30.5 gallon wing fuel tank sight gages | | F. Atlee Dodge Master Data List No. 3286MDL Revision A, dated 16 January 2004, or later approved Revisions. FAA STC SA02115AK dated 9 June 1998, revised on 20 January 2004. 3286IM -Installation and Calibration Procedures | 1. Any Piper aircraft incorporating this change must be equipped with F. Atlee Dodge 30.5 gallon wing fuel tanks in accordance with FAA STC SA649AL, SA00092SE, or SA5781NM.2. This STC is approved only for the product configuration as defined in the app (...) | 09.06.2005 |
| 10042946 | F. ATLEE DODGE AIRCRAFT SERVICES | EASA.IM.A.S.01073 | | FAA STC SA00092SA | PA-22-150, PA-22-160 | 30.5 Gallon Fuel Tanks manufactured in accordance with Drawing List 30.5GAL TANKS, dated 09/03/1992, and installed in accordance with Installation Instructions No 3258-1, rev 1, dated 10/02/1994 or later approved revisions | | -FAA STC No SA00092SA-Installation Instructions No 3258-1-FAA approved Supplemental Flight Manual, dated 16/10/1998, or later FAA approved Revision | 1. FAA approved Supplemental Flight Manual, dated 16 October 1998, or later approved revision is a required part of this installation. This STC is approved only for the product configuration as defined in the approved design data referred to in the pa (...) | 22.08.2005 |
| 10016731 | FARGO MANUFACTURING | EASA.IM.R.S.01055 | | FAA STC SR01926CH | MD HELICOPTERS MD900 | Install 61.8 gallon crashworthy auxiliary fuel tank on the aft cabin baggage bay floor | | Fargo Installation Instructions A-900-3CW, Revision C dated January 08 2004 or later approved revision & Fargo Manufacturing Drawing List A-900-3CW Revision 2 dated October 14 2003 or later approved revision. Fargo Manufacturing RFM Supplement No. Rev (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 30.03.2005 |
| 10017067 | FARGO MANUFACTURING | EASA.IM.R.S.01503 | | FAA STC SR01813CH | MD HELICOPTERS MD900 | Installation of Auxiliary Fuel System Kit in accordance with FAA STC SR01813CH | | EASA approved Flight Manual Supplement, EASA Doc No. 900, dated 24.11.2009 or later EASA approved revision Fargo Installation Instructions, Doc No. A-900-224, Rev A, dated 24.04.2008 or later EASA approved revision Fargo Manufacturing Drawing List, Do (...) | This STC is limited to the installation of Fargo Manufacturing Co. 52.5-gallon crashworthy auxiliary fuel tank pursuant to FAA STCSR01238CH as referred in FAA STC SR01813CH. The 55-gallon installation and the 28-gallon installation are not included in this STC. | 15.12.2008 |
| 10015060 | FEDERATION FRANÇAISE DE VOL A VOILE | EASA.A.S.03383 | | | | Winglet Retrofitting. | | Dossier de modification Supplement aux Manuels de vol Statement of conformity. | 2.2 Supplemental Flight Manual | 01.02.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|--------------------------------|--|--|---|--|--|---------------|
| 10030091 | FEDERATION FRANCAISE DE VOL A VOILE | | | | ASW20F | STC M-01/09-A01 Winglet Installation on ASW20F | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Dossier STC M-01/09-A01 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 20.05.2010 |
| 10043603 | FEDERATION FRANCAISE DE VOL A VOILE | | | | 101 | Polyurethane paint - M02-12-A01 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions level of the original product are unchanged and remain applicable to this certificate/ approval. | Supplément au Manuel d'Entretien C101 PEGASE Rev 1-1.doc; or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 12.02.2013 |
| 10043604 | FEDERATION FRANCAISE DE VOL A VOILE | | | | ASW 20F, ASW 20 FL | Polyurethane paint - M03-12-A01 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions level of the original product are unchanged and remain applicable to this certificate/ approval. | Supplément au Manuel d'Entretien ASW 20 F FL Rev 1or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 12.02.2013 |
| 10044028 | FEDERATION FRANCAISE DE VOL A VOILE | | | | STANDARD CIRBUS CS 11-75L | Winglets Retro Fitting - CS 11/75 L | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | SUPPLEMENT AU MANUEL DE VOL CIRBUS... doc; or later revisions of the above listed documents approved by EASA | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 13.03.2013 |
| 10032277 | FIBERGLAS-TECHNIK RUDOLF LINDNER | | | | GROB G 103 "TWIN II" | STC "Reparatur Twin II" | The Certification Basis for the original product remains applicable to this certificate/ approval. | STC "Reparatur Twin II" - Flight manual prepared for the S/N of the fuselage and fitted with "Twin II" - Flughandbuch zusammengestellt für die Werknummer des Rumpfes und passend zum "Twin II". Maintenance manual prepared for the S/N of the wing and f (...) | Even if parts from a "Twin II" "Acro" are used, the combined airplane will be always a "Twin II" (without Acro). / Auch wenn Teile eines "Twin II" "Acro" verwendet werden, das neu entstandene Segelflugzeug ist immer ein "Twin II" (ohne Acro). | 22.10.2010 |
| 10015983 | FIELD AVIATION COMPANY INC. | EASA.IM.A.S.02135 | | SA93-81 | BOMBARDIER DHC-6 BOMBARDIER DHC-6 SERIES 200 BOMBARDIER DHC-6 SERIES 300 | Roof Access Steps Installation | | Transport Canada SA93-81 Field Aviation Company Inc Engineering Data Summary 112200 Revision A dated April 13, 1993 or alter approved revision | -- | 13.11.2006 |
| 10016003 | FIELD AVIATION COMPANY INC. | EASA.IM.A.S.02164 | | O-LSA06-179/D ISSUE 1 (TCCA LI | BEECH B200 SUPER KING AIR | Installation of a single Universal UNS-1F Flight Management System complete with dual Universal GLS-1250 DGPS Receivers (SCAT-1). Transport Canada LSTC O-LSA06-179/D Issue 2. | | Configuration AUNS-1F c/w GLS-1250 Electrical and structural Provisions:- Installation of the UNS-1F retrofit modifications must be performed in accordance with FAEL Master Drawing and Data List designated DDL-1163 Rev IR, or later approved revisions. (...) | Aircraft Serial BB-1547 only The UNS-1F Flight Management System is approved for En-route, Terminal and Non Precision Approach Operations only. | 05.02.2007 |
| 10016129 | FIELD AVIATION COMPANY INC. | EASA.IM.A.S.02323 | REV. 1 | TCCA STC SA07-80 | BOMBARDIER DHC-8-311 | Retrofit of Swedish Coastguard Maritime Patrol Modification. TCCA STC Number: SA07-80, Issue 2 dated 26 February 2008 | | Installation must be performed in accordance with FAEL Master Drawing and Data List designated DDL-1141 Rev S. Flight Operations must be performed in accordance with FAEL Flight Manual Supplements as follows: FMS 20113 Rev A Dual UNS-1F Flight Mana (...) | Limitations UNS-1F Flight Management Systems approved for En-route, Terminal and Non Precision approach operations only. External Mission LAN System not approved for airborne operations. External Nokia Cellular Telephone System not approved for airborne (...) | 01.07.2008 |
| 10016225 | FIELD AVIATION COMPANY INC. | EASA.IM.A.S.02476 | | TCCA C-LSA07-021/D | BOMBARDIER CL-600-2B19 | Installation of Corporate Shuttle Interior. | | STC C-LSA07-021/D, Issue 1 dated 16 September 2007. Installation of the corporate shuttle interior must be in accordance with Transport Canada Civil Aviation (TCCA), approved Field Aviation Company Inc. (FACI) Modification instruction 204910, Revision (...) | Limited to serial no. 8049 | 09.10.2007 |
| 10016299 | FIELD AVIATION COMPANY INC. | EASA.IM.A.S.02596 | | TCCA STC C-LSA07-322/D | BOMBARDIER CL-600-2B19 | Installation of Life Rafts. | | Field Aviation Company Inc. (FACI) Modification Instructions 213610, Revision - original issue-, dated September 7, 2007 or later approved revision. TCCA LSTC No. C-LSA07-322/D | Aircraft Serial No. 8049 only | 12.10.2007 |
| 10016417 | FIELD AVIATION COMPANY INC. | EASA.IM.A.S.02750 | | TCCA STC SA05-73 | BOMBARDIER DHC-8-311 | Installation of a Artex C406-2 or C406-N Three Frequency Emergency Locator Transmitter (ELT) | | Maintenance Manual Supplement 200331. | | 11.06.2008 |
| 10016418 | FIELD AVIATION COMPANY INC. | EASA.IM.A.S.02751 | | TCCA STC SA00-2 | BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-300 SERIES BOMBARDIER DHC-8-311 BOMBARDIER DHC-8-315 | Ultra Electronics ANVC-Active Noise and Vibration Control System and PBMS # Propeller Balancing and Monitoring System Ref. TCCA STC SA00-2, Issue 4, dated 24 October 2004 | | Configuration A - Ultra Electronics ANVC - Active Noise and Vibration Control System, fitted with a Post 1990 Heath Techna Interior (Structural and Electrical Provisions Only) PBMS - Propeller Balancing and Monitoring System (Structural and Electrical P (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 25.08.2008 |
| 10016419 | FIELD AVIATION COMPANY INC. | EASA.IM.A.S.02752 | | TCCA LSTC C-LSA06-101/D | BOMBARDIER DHC-8-311 | Installation of a 320cft baggage compartment and installation of PBE's and O2 bottle into the cabin (Ref. TCCA LSTC C-LSA06-010/D) | | Installation of 320cft baggage compartment must be performed in accordance with Modification Instruction 202410*) Installation of O2 bottle and PBE into the cabin must be performed in accordance with Modification Instruction 20261*) Maintenance must be performed in accordance with DHC-8-300 Main (...) | | 06.10.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|-------------------------|----------------|---------------------------|---|---|---|---|---|---------------|
| 10016430 | FIELD AVIATION COMPANY Inc. | EASA.IM.A.S.02769 | REV. 1 | TCCA STC C-LSA08-056/D | DHC-8-202, DHC-8-311 | Honeywell AHRS to SFIM APIRS Conversion Provisions TCCA STC C-LSA08-056/D, Issue 2 | FAR 25 as per EASA.IM.A.191, Issue 5 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions le (...) | Installation of the Conversion Provisions for Honeywell AHRS to SFIM APIRS Retrofit must be in accordance with Transport Canada Civil Aviation (TCCA) approved Field Aviation Company Inc. (FACI) Modification Instructions 215710, Revision B, dated Nov 7, 20 (...) | Installation is limited to Serial Numbers 435, 440, 441 and 407 | 23.05.2012 |
| 10016432 | FIELD AVIATION COMPANY INC. | EASA.IM.A.S.02772 | | TCCA STC O-LSA08-023/D | BOMBARDIER DHC-8-311 | VHF Comm Replacement for 8.33 kHz Spacing, ref. TCCA LST C-LSA08-023/D. | | Installation law modification instruction 215610. Maintenance must be performed in accordance with Maintenance Manual Supplement 215631. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 15.08.2008 |
| 10016603 | FIELD AVIATION COMPANY INC. | EASA.IM.A.S.03017 | | TCCA STC SA01-55 | BOMBARDIER DHC-8-400 SERIES BOMBARDIER DHC-8-401 BOMBARDIER DHC-8-402 | Installation of Variable Position Cabin Class Divider. | | 1. Bin Mount Type :- Installation in accordance with Field Aviation Company Inc. Drawing 17342505, Original Issue, dated 5 April 2001 *); - Maintenance must be performed in accordance with Field Aviation Company Inc. Component Maintenance Ma (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 28.08.2008 |
| 10016648 | FIELD AVIATION COMPANY INC. | EASA.IM.A.S.03092 | REV. 1 | TCCA (LSTC) C-LSA08-018/D | BOMBARDIER DHC-8-401, -402 | G3 Water System Modification: TCCA LSTC C-LSA08-018/D, Issue 3, dated 14 September 2009. | See TCDS EASA.IM.A.191 The Certification Basis for the original product remains applicable to this certificate. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | G3 Galley modification and under floor plumbing must be in accordance with TCCA approved FACI Modification Instruction, 213011, dated 15 October 2008 *). Instructions for Continuing Airworthiness Maintenance must be in accordance with TCCA approved FACI (...) | None. | 21.09.2009 |
| 10027252 | FIELD AVIATION COMPANY Inc. | P- EASA.IM.A.S.02934 | | TCCA STC SA09-15 | DHC-8-314 | Icelandic Coast Guard Maritime Patrol Modification TCCA STC Number: SA09-15, Issue 1 dated 29. April 2009 | FAR 25 as per TCDS EASA.IM.A.191 plus EASA Special Condition "Service Compartment" as defined in CRI D-02, Issue 2, Annex 1. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are appl (...) | Installation must be performed in accordance with FAEL Master Drawing Data List designated DDL-1185, Rev C Flight Operations must be performed in accordance with FAEL Flight Manual Supplements as follows: FMS 20164 Rev IR, Mission Landing Gear Warni (...) | UNS-1F(w) Flight Management Systems approved for En-route, Terminal and Non Precision approach operations only. External LAN System not approved for airborne operations. External Nokia Cellular Telephone System not approved for airborne use. The authori (...) | 16.09.2009 |
| 10029315 | FIELD AVIATION COMPANY INC. | | | TCCA STC SA02-86 | DHC-8-102, -103, -106, -201 DHC-8-202, -301, -311, -314 DHC-8-315 | Installation of a HF Antenna into the Vertical Stabilizer Leading Edge for an Allied Signal KHf-950 HF Transceiver. TCCA STC SA02-86, Issue 2 | FAR-25 as per EASA.IM.A.191. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Special Condition H-01 "Enhanced Airworthiness Programme for (...) | Installation must be in accordance with Field Aviation Company Inc. Modification Instructions 178010, Rev. A. Maintenance must be in accordance with Field Aviation Company Inc. Maintenance Manual Supplement 178031, dated 15 January 2002 or later revision (...) | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 18.03.2010 |
| 10030757 | FIELD AVIATION COMPANY Inc. | EASA.IM.A.S.03226 | REV. 1 | TCCA STC SA92-23 | DHC-8-102/-103/-106/-201/-202 DHC-8-301/-311/-314/-315 | De-icer Boot Installation | FAR Part 25 as per TCDS EASA.IM.A.191 for DHC-8-100/-200/-300 Series | Installation of Aerazur Type 51 De-Ice Boots in accordance with Fields Aviation Company Inc. Drawing List No. 33644; Issue 2. Instructions for Continued Airworthiness Maintenance must be done in accordance with Fields Aviation Company Inc. Maintenance Man (...) | None. | 28.05.2013 |
| 10031816 | FIELD AVIATION COMPANY Inc. | CAA STC 2004-5141 | | TCCA STC SH88-1 | AS 332 L, AS 332 L1 | Installation of 4 Place Auxiliary Seat Tank This EASA STC revises CAA STC 2004-5141 dated 14 May 2004 and adds model AS332L1. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Tank Installation 300151-3-6346 revision D dated 14 October 1987 - Drawing List; E300151 revision G dated September 1987 - Master Drawing; -FACI-6346-01 amendment 1 dated 6 November 1987 - Rotorcraft Flight Manual Supplement; -FACI-6346-02 amendment 1 (...) | None | 22.12.2010 |
| 10032965 | FIELD AVIATION COMPANY INC. | | | TCCA LSTC C-LSA09-212/D | DHC-8-311 | Installation of Aft Baggage Bay Divider Nets TCCA LSTC C-LSA09-212/D | FAR 25 as per EASA.IM.A.191 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original p (...) | Installation must be in accordance with TCCA approved Field Aviation Company Inc. Modification Instruction 218110, IR, dated 30 June 2009. Maintenance must be in accordance with TCCA accepted Field Aviation Company Inc. Maintenance Manual Supplement 218 (...) | The modification is applicable only when the necessary structural provisions are pre-existing. | 10.12.2010 |
| 10035887 | FIELD AVIATION COMPANY Inc. | | REV. 1 | | DHC-8-201/-202 DHC-8-102/-103/-106 DHC-8-301/-302 | Auxiliary Power Unit (APU) Removal TCCA STC SA98-30, Issue 3 | FAR 25 as per EASA.IM.A.191, Issue 4 (incl. SC H-01 ICA on EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ o (...) | APU removal in the DHC-8-100 and 200 series must be done in accordance with TCCA approved Field Aviation Company Inc. Engineering Order 141103, Revision G, dated 29 June 2005. APU removal in the DHC-8-300 series must be done in accordance with TCCA ap (...) | The modification is applicable only when the necessary structural provisions are pre-existing | 28.07.2011 |
| 10035939 | FIELD AVIATION COMPANY Inc. | | | | DHC-8-102/-103/-106 | Installation of Right Hand Ground Air Connection TCCA STC C-LSA11-043/D, Issue 1 | FAR 25 as per EASA.IM.A.191, Issue 4 (incl. SC H-01 ICA on EWIS) The approval holder shall fulfil the obligations of Part 21, Paragraph 21.A.109. The Certification Basis for the original product remains applicable to this certificate/ approval. The requir (...) | Installation of the Right Hand Ground Air Connection must be in accordance with Transport Canada Civil Aviation (TCCA) approved Field Aviation Modification Instruction 218210, Revision IR, dated July 21, 2010. Maintenance of the Ground Air System is unc (...) | OEM installed APU must be removed in accordance with EASA STC 10035887 or TCCA STC SA98-30, Issue 3. | 29.07.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|-----------------------|----------------|------------------------|--|---|--|---|--|---------------|
| 10036488 | FIELD AVIATION COMPANY Inc. | | | TCCA STC NO. SA11-37 | DHC-8-401, -402 | Temporary Removal of Auxiliary Power Unit (APU) TCCA STC SA11-37, Issue 1 | CS/JAR-25 as per EASA.I.M.A.191, Issue 4 (incl. SC H-01 ICA on EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions le (...) | The temporary APU removal series must be done in accordance with TCCA approved Field Aviation Company Inc. Modification Instruction 25341000, Revision A, dated 02 June 2011. Maintenance of the APU System is unchanged from the instructions provided in Bom (...) | The modification is applicable only when the necessary structural provisions are pre-existing | 13.09.2011 |
| 10038998 | FIELD AVIATION COMPANY Inc. | | REV. 1 | TCCA STC O-LSA06-139/D | DHC-8-106 DHC-8-202 | TCCA STC O-LSA06-139/D, Issue 2 Retrofit installation of a Honeywell KHF-950 HF Radio System. | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions le (...) | Installation must be in accordance with TCCA approved Field Aviation East Ltd. Master Drawing and Data List DDL-1158, Rev. C dated 15 August 2011. Maintenance must be in accordance with TCCA accepted Maintenance Manual Supplement 20169, Rev. A, dated 08 (...) | The modification is limited to Serial numbers 335, 337, 445, 446. | 19.01.2012 |
| 10038104 | FIELD AVIATION COMPANY Inc. | | | TCCA STC O-LSA06-111/D | DHC-8-202 | TCCA STC O-LSA06-111/D, Issue 2 Retrofit of AirCall ST3100 Satellite Telephone and Tracking System. | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions le (...) | Installation must be in accordance with TCCA approved Field Aviation East Ltd. Master Drawing and Data List DDL-1155, Rev. C, dated 22 July 2011. Maintenance must be in accordance with TCCA accepted Maintenance Manual Supplement 20162, Rev. B, dated 22 Ju (...) | The modification is limited to Serial numbers 445, 446. | 19.01.2012 |
| 10038661 | FIELD AVIATION COMPANY Inc. | | | TCCA STC O-LSA11-232/D | DHC-8-103 | Honeywell Mark VII EGPWS Annunciator Relocation and Upgrade TCCA STC O-LSA11-232/D, Issue 1 | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions le (...) | Installation must be in accordance with TCCA approved Field Aviation Company Inc. Master Drawing and Data List designated 25550000, Rev. IR. Flight operations must be conducted in accordance with FACI Flight Manual Supplement designated FMS 25553000 Rev (...) | This modification is limited to Serial number 310 with EASASTC No.MOT.21NE2.138 installed. | 06.03.2012 |
| 10038932 | FIELD AVIATION COMPANY Inc. | | | | DHC-8-102/-103/-106/-201/-202 DHC-8-301/-311/-314/-315 | Auxiliary Wing Fuel Tank Installation TCCA STC SA11-64, Issue 2 | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. SC H-01) The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Special Condition CRI E-01 (Fuel T (...) | DHC-8-100 and 200 Series Installation must be performed in accordance with FACI Master Drawing and Data List designated 25370000, Revision B. Flight Operations must be performed in accordance with FACI Flight Manual Supplement FMS 20182, Revision C. Inst (...) | None | 23.03.2012 |
| 10038999 | FIELD AVIATION COMPANY Inc. | JAA STC ACG.21.NE.050 | | TCCA STC SA01-50 | DHC-8-102/-103/-106 DHC-8-201/-202 DHC-8-301/-311/-314/-315 | Installation of Variable Position Cabin Class Divider TCCA STC SA01-50, Issue 2 | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions le (...) | DHC-8-100/ 200 Series Aircraft:- Installation of the Variable Position Cabin Class Divider must be in accordance with TCCA approved Field Aviation Company Inc. Modification Instruction 173311, Rev. IR, dated 25 July 2011.-Maintenance of the Variable Po (...) | None | 29.03.2012 |
| 10039339 | FIELD AVIATION COMPANY Inc. | | REV. 1 | TCCA STC SA11-27 | DHC-8-102, -103, -106, DHC-8-201, -202 DHC-8-301, -311, -314, -315 | Universal EFI-890R EFIS Cockpit Upgrade TCCA STC SA11-27, Issue 3 | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. SC H-01) The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. FAR 25.1316 amendment 25-134 FAR 25 (...) | DHC-8-100 and 200 Series Installation must be performed in accordance with FACI Master Drawing and Data List designated 25200000 Rev E. Flight Operations must be performed in accordance with FACI Flight Manual Supplements listed as follows: 25203000 Rev (...) | The installation of the DHC-8-314 MPA configuration is limited to Serial number 0660. The installation of dual, Universal UNS-1E, or UNS-1E(w) or UNS-1F or UNS-1F(w) Flight Management Systems are a prerequisite to the installation of the EFI 890R Electronic Flight Instrument Suite. | 03.05.2012 |
| 10039340 | FIELD AVIATION COMPANY Inc. | | REV. 1 | TCCA STC C-LSA12-014/D | DHC-8-202 | Installation of upgraded Cabin Floor Panels TCCA STC C-LSA12-014/D, Issue 1 | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions le (...) | Installation must be in accordance with TCCA approved Field Aviation Company Inc. Master Data List 25500000, Rev. A), dated 23 April 2012. Maintenance is unchanged from the instructions provided in the Bombardier DHC-8-200 Maintenance Manual PSM 1-62- (...) | The modification is limited to Serial numbers 445, 446 without 1990 Hunting interiors only. Floor loads of 75 lbs/ft(s) only in conjunction with underfloor support structure modified per Bombardier CR 853CH2003 or equivalent. | 03.05.2012 |
| 10039731 | FIELD AVIATION COMPANY Inc. | | | TCCA STC C-LSA12-017/D | DHC-8-202 | Fleet Standardization Modification TCCA STC C-LSA12-017/D, Issue 1 | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. EWIS). The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions le (...) | Installation must be in accordance with TCCA approved Field Aviation Master Data List 25440000, Rev. IR, dated 26 March 2012. Maintenance of the Fleet Standardization Modification Installation must be in accordance with Field Aviation Maintenance Manual (...) | The modification is limited to Serial numbers 435, 440 and 441. | 21.05.2012 |
| 10039983 | FIELD AVIATION COMPANY Inc. | | | TCCA STC C-LSA12-006/D | DHC-8-202 | RH Rear Facing Seat Installation. TCCA STC C-LSA12-006/D, Issue 1. | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions le (...) | Installation must be in accordance with TCCA approved Field Aviation Master Data List 25410000, Revision A. Maintenance must be in accordance with TCCA accepted Field Aviation Aircraft Maintenance Manual Supplement 25413100, Revision IR or later revisio (...) | The modification is limited to Serial numbers 435, 440 and 441 with post-1990 Hunting or C->-&-<->D interiors only,... | 05.06.2012 |
| 10041560 | FIELD AVIATION COMPANY Inc. | | | TCCA STC C-LSA12-128/D | DHC-8-202 | Auxiliary Ground Heater System TCCA STC LSA12-128/D, Issue 1 | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions le (...) | Installation must be in accordance with Field Aviation Master Data List 25390000, Revision IR. Maintenance must be in accordance with Field Aviation Aircraft Maintenance Manual Supplement 25393100, Revision IR or later revisions of the above listed docum (...) | Installation is limited to Serial Numbers 435, 440, 441, 445 and 446 | 26.09.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|----------------------|----------------|------------------------|---|--|--|---|--|---------------|
| 10041590 | FIELD AVIATION COMPANY Inc. | | REV. 1 | TCCA STC SA-12-57 | DHC-8-102, DHC-8-103, DHC-8-106. | Installation of Protective Breathing Equipment (PBE)TCCA STC SA 12-57, Issue 1 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation must be in accordance with Field Aviation Master Data List 25680000, Revision IR, dated 05 September 2012; Maintenance must be in accordance with Field Aviation Aircraft Maintenance Manual Supplement 25683100, Revision IR, dated 27 August 2 (...) | None | 15.10.2012 |
| 10046157 | FIELD AVIATION COMPANY Inc. | | | TCCA SA99-261, ISSUE 3 | DHC-8-102, DHC-8-103 DHC-8-106 DHC-8-201, DHC-8-202 DHC-8-301, DHC-8-311 DHC-8-314, DHC-8-315 | Installation of Aerazur Engine Inlet De-Icer Boot. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation of Aerazur Engine Inlet De-Icer Boot P/N 51SP162 31-1 in accordance with Aerazur Component Maintenance Manual for P/N 51SP16231-1, Revision 2. Instructions for Continued Airworthiness Maintenance in accordance with respective DHC-8 Maintenance (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 22.08.2013 |
| 10015754 | FIELD AVIATION EAST, LTD. | EASA.I.M.A.S.01787 | REV 2 | SA04-125 ISSUE 1 | BOMBARDIER DHC-8-102 BOMBARDIER DHC-8-103 BOMBARDIER DHC-8-311 BOMBARDIER DHC-8-315 | Collins CTL-94E Controller for Elementary Surveillance | | Master Drawing & Data List DDL-1128 | | 26.02.2007 |
| 10015755 | FIELD AVIATION EAST, LTD. | EASA.I.M.A.S.01788 | REV 1 | SA05-67 ISSUE 1 | BOMBARDIER DHC-8-102 BOMBARDIER DHC-8-103 BOMBARDIER DHC-8-311 BOMBARDIER DHC-8-315 | Installation of ICARUS Sky Connect Tracker SOP and Satellite Communication System | | - FAEL Master Drawing List DDL-1140 Rev. A or later Transport Canada approved revision. - FAEL Maintenance Manual Supplement MMS 20148 Revision IR or later Transport Canada accepted revision. | This approval is only applicable to the type/model of aeronautical product specified therein. Prior to incorporating this modification, the installer shall establish that the interrelationship between this change and any other modification(s) incorporated (...) | 04.08.2006 |
| 10015849 | FIELD AVIATION EAST, LTD. | EASA.I.M.A.S.01949 | REV. 2 | TCCA STC SA05-42 | DHC-8-102, -103, -106 DHC-8-201/-202 | Installation of Dual Universal GLS-1250 DGPS Receivers (SCAT-1)TCCA STC SA05-42, Issue 3 | FAR Part 25 as per TCDS EASA.I.M.A.191, Issue 5 (incl. SC H-01)The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Configuration A - (Electrical and Structural Provisions only) Installation must be performed in accordance with FAEL Master Drawing and Data List designated DDL-1136 Revision A. Instructions for Continuing Airworthiness Maintenance must be performed in (...) | FAEL (Dual UNS-1F FMS for Software versions 801/802 and 803) STCEASA.I.M.A.S.3163, Issue 1 or later EASA approved revision must be installed as a prerequisite to the installation of the Dual GLS 1250DGPS (SCAT-1) modification. | 20.01.2012 |
| 10027122 | FIELD AVIATION EAST, LTD. | P-EASA.I.M.A.S.03183 | REV. 2 | TCCA STC SA02-48 | DHC-8-102, -103, -106 DHC-8-201/-202 DHC-8-301, -311, -314, -315 | Retrofit Installation of Universal Avionics Systems Corp. UNS-1F FMS for Software versions SCN 801/802 and 803Ref. TCCA SA02-48, Issue 5 | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. SC H-01)The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Config. A - EFIS Equipped Aircraft, Single UNS-1F (Provisions Only)DHC-8-100 Series: Installation must be performed in accordance with Field Aviation Master Drawing and Data List designated DDL 1073, Rev. F) DHC-8-300 Series: Installation must be performed in (...) | Configuration B Single UNS-1F with Software Version SCN 801, EFIS Equipped Aircraft - de Havilland factory installed single UNS-1B or UNS-1B+ Flight Management System Electrical and Structural provisions are a prerequisite to the Universal UNS-1F Flight Ma (...) | 31.01.2012 |
| 10038517 | FIELD AVIATION EAST, LTD. | | | TCCA STC SA05-8 | DHC-8-202 | Max. Gross Takeoff Weight Increase to 37.300 lbsTCCA STC SA05-08, Issue 1 | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. SC H-01)The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation must be performed in accordance with FAEL Master Drawing and Data List designated DDL1124, Revision H. Flight Operations must be performed in accordance with FAEL Flight Manual Supplement designated FMS 20083, Revision B. Instructions for C (...) | None | 22.02.2012 |
| 10038533 | FIELD AVIATION EAST, LTD. | | | TCCA STC O-LSA11-177/D | DHC-8-202 | Cabin Drop Down Oxygen System Removal and Cabin Interior RefurbishmentTCCA STC O-LSA11-177/D, Issue 1 | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. EWIS)The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation must be in accordance with TCCA approved Field Aviation Master Drawing and Data List 25470000, Revision A. Maintenance must be in accordance with TCCA accepted Field Aviation Maintenance Manual Supplement 25473100, Revision IR or later rev (...) | The modification is limited to Serial numbers 445, 446. | 23.02.2012 |
| 10039284 | FIELD AVIATION EAST, LTD. | | REV. 1 | TCCA STC O-LSA11-147/D | DHC-8-202 | Air Iceland Avionics UpgradeTCCA STC O-LSA11-147/D, Issue 1 | FAR 25 as per EASA.I.M.A.191, Issue 5 (incl. EWIS)The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation must be in accordance with TCCA approved FAEL Master Data List designated 25380000, Revision C. Operation must be in accordance with TCCA approved FAEL Airplane Flight Manual Supplements as follows: -25383000 Revision IR - TCAS II, Change 7 (...) | The modification is limited to Serial numbers 445, 446. Universal UNS-1E(w) Flight Management Systems are approved for IFR en-route, Terminal and Non Precision approach operations. | 03.05.2012 |
| 10043261 | FIELD AVIATION EAST, LTD. | | | TCCA STC O-LSA12-130/D | DHC-8-202 | Dual Universal UNS-1E(w) Flight Management System Upgrade | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation must be performed in accordance with FACI Master Drawing and Data List N° 25670000, Rev. IR<[>->]. Operations must be performed in accordance with FACI Flight Manual Supplement N° 25673000, Rev IR<[>->]. Maintenance must be performed in (...) | Limited to Serial Numbers 477, 488, 496, 498 and 504 with EASA STC10028595 (Advanced Avionics Technology) installed. | 16.01.2013 |
| 10046177 | FIELD AVIATION EAST, LTD. | | REV. 1 | TCCA SA13-36, ISSUE 1 | DHC-8-102, DHC-8-103 DHC-8-106 DHC-8-201, DHC-8-202 DHC-8-301, DHC-8-311 DHC-8-314, DHC-8-315 | Flight Crew, Zodiac MF20-517 Full Face Oxygen/ Smoke Mask Upgrade. Rev. 1 was issued to correct a mistake in the STC holder name. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation must be in accordance with TCCA approved FACI Master Drawing List designated 25780000, Revision A. Instructions for Continued Airworthiness Maintenance must be performed in accordance with TCCA accepted FACI Maintenance Manual Supplement d (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 29.08.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|-------------------|----------------|-----------------------|--|---|---|--|---|---------------|
| 10017035 | FILTRATION DEVELOPMENT CORPORATION | EASA.IM.R.S.01459 | | FAA STC SR01616SE | SIKORSKY S-76A SIKORSKY S-76A++ SIKORSKY S-76C SIKORSKY S-76C+ | Installation of an Engine Air Filter System for Sikorsky S-76 models, A+, A++ and C & C+ | | - FDC/aerofilter Rotorcraft Flight Manual Supplement 1076-1200-105 (for S-76A+/A++/C) - FDC/aerofilter Rotorcraft Flight Manual Supplement 1076-1200-101 (for S-76C+) - FDC/aerofilter 'Documentation Package' 1076-SERIES-DPG-1 Rev IRor subseq (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 05.02.2009 |
| 10014229 | FINNAIR OYJ | EASA.A.S.01392 | | | BOEING MD 11 | MD11 Cabin Crew Rest Area | | Data package containing the following documents, and additional supporting data:1. MD-11 C/A Bunk Airworthiness Compliance Sheet Rev. E2. Cabin Crew Rest Compartment Modifications3. MD-11 C/A Bunk AYM60 and AYM61 Flight Test Report Rev. B4 (...) | Emergency evacuation procedures, including the emergency evacuation of an incapacitated occupant from the Crew Rest Area, must be established. All of these procedures must be transmitted to the operator for incorporation into their training programs (...) | 15.08.2006 |
| 10016090 | FIREWALL FORWARD TECHNOLOGIES LLC | EASA.IM.A.S.02277 | | FAA STC SA01435LA | | Installation of engines, modified with a Firewall Forward Technologies, LLC modified camshaft, into aircraft listed on the EASA approved ATL | | FAA approved Firewall Forward Technologies, LLC Installation Instructions FFT-INI-001-AFR, revision -, dated 06-Feb-2003, or later approved revision.Continued Airworthiness Instructions, document number FFT-CAI-001-AFR, revision A, dated 11-Sep-2003 or later released revision | 1. This installation must be done concurrently with EASASTC No. EASA.IM.E.S.01010, which modifies the engine.2. Regardless of the EASA Approved Type List (ATL), installation is only approved if the Airplane Type / Variant itself is EASA approved (...) | 12.02.2007 |
| 10016691 | FIREWALL FORWARD TECHNOLOGIES LLC | EASA.IM.E.S.01005 | | FAA STC SE00584DE | | Installation of modified Oil Filter Adapter, an increased opening pressure spring for filter bypass, and modified filter stud in accordance with Firewall Forward Technologies, Inc. Master Document List No. FFT-MDOCLST-OILSTUD-001, Rev. B, dated August 8, 2005 or later FAA approved revision. | | Installation Instructions FFT-INLYC-OILSTUD-001Continued Airworthiness Instructions FFT-CAIYC-OILSTUD-001 | Compatibility of this design change with previously approved modifications must be determined by the installer.If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written (...) | 18.01.2007 |
| 10016696 | FIREWALL FORWARD TECHNOLOGIES LLC | EASA.IM.E.S.01010 | | FAA STC SE01436LA | | Installation of Firewall Forward Technologies, LLC modifiedcamshaft in accordance with Master Drawing List FWF-MDL-ST9721 LA-E, revision G, dated 2/14/03 or later FAA approved revision. | | Installation Instructions FFT-INI-001Continued Airworthiness Instructions FFT-CAI-001 | This installation must be done concurrently with EASA STC No. EASA.IM.A.S.02277, which modifies the airplane installation.Compatibility of this design change with previously approved modifications must be determined by the installer. The installatio (...) | 06.02.2007 |
| 10016697 | FIREWALL FORWARD TECHNOLOGIES LLC | EASA.IM.E.S.01011 | REV. 1 | FAA STC SE00590DE | | Installation of a Lubricated Camshaft in accordance with Master Drawing List FFT-MDL-ST3370DE-E, Revision IR dated September 9, 2005, FAA approved September 26, 2005, or later FAAapproved revision. | | Installation Instructions FFT-INI-002Continued Airworthiness Instructions FFT-CAI-002 | 1. All engine installation criteria and limitations are unchanged and remain in effect. Installation of engines modified by this STC must be in accordance with the original aircraft installation criteria.2. Instructions for Continued Airworthiness, (...) | 05.02.2007 |
| 10015970 | FIRST AIR MAINTENANCE SERVICES | EASA.IM.A.S.02121 | | SA06-39 | AEROSPATIALE ATR 42-300 AEROSPATIALE ATR 42-320 AEROSPATIALE ATR 72-101 AEROSPATIALE ATR 72-102 AEROSPATIALE ATR 72-201 AEROSPATIALE ATR 72-202 | Installation of Dual SAGEM F227 APIRS Attitude and Heading Reference System | | Installation must be in accordance with First Air Master Data List MDL-06-001, Revision A, dated 20th November 2006 or later approved revisions.Operation must be in accordance with First Air Flight Manual Supplement AFMS 06-004 issue 1, 13th March 2 (...) | This STC is approved only for the aircraft modelsATR 42-200/ 300 / 320 / 400 / 500ATR 72-101 / 102 / 201 / 202 / 211 / 212 / 212 Afor those serial numbers fitted with the Honeywell AH600 AHRS as per originally manufactured and delivered by ATR (Type (...) | 16.01.2007 |
| 10017203 | FISCHER + ENTWICKLUNGEN | EASA.R.S.01352 | | | BELL 427 | Installation of floor mounted F+E Medical Seats 230/305 into Bell 427 cabin by use of installation racks attached to forward and aft cabin bulkhead fittings | | Flight Manual Supplement D0704-021 revision - dated 17 July2007 or later approved revisionMaintenance Manual Supplement D0704 - 020 revision 0 dated 30 March 2007 or later approved revisionInstallation Instructions are contained in the Maintenance M (...) | | 20.07.2007 |
| 10015484 | FLIGHT ENVIRONMENTS INC | EASA.IM.A.S.01244 | | FAA STC SA01318LA | BEECH 390 | Installation of thermal/acoustic insulation kit | | -FAA STC No SA01318LA-MM-Supplement 0219.ICA, Rev B or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 26.07.2005 |
| 10015709 | FLIGHT STRUCTURES, Inc. | EASA.IM.A.S.01702 | | FAA STC ST02944AT | B777-200, B777-300 | Installation of Cabin Wireless Local Area Network (CWLAN) Distribution System | | MDL: R44-0242, Revision Y, dated 19th April 2006 or later FAA approved revisionAFM Supplement: R44-0316, Revision D, dated 20th June 2006 or later FAA approved revision | Airworthiness Limitations Section of the instruction for Continuous Airworthiness, document numbers F44-0246, revision A, dated 20thOctober 2004 or R44-0406, revision IR, dated 23rd June 2005 or later FAA approved revisions | 20.07.2006 |
| 10015719 | FLIGHT STRUCTURES, Inc. | EASA.IM.A.S.01717 | | FAA STC ST01729SE | B747-400 | Business Class Seat Installation | | Flight Structures Inc Master Drawing List A5FS513-D01 Revision D dated January 04 2007 or later approved revision. | Boeing Service Bulletin 747-25-3420 dated June 13 2006 mustbe installed concurrently with this STC. | 17.01.2007 |
| 10015720 | FLIGHT STRUCTURES, Inc. | EASA.IM.A.S.01719 | REV. 3 | FAA STC NO. ST01849SE | 777-200 | Business Class Seat Installation | IM.A.003The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalCRI-H01 EWISThe requirements for environmental protection and the associatedcer (...) | FAA STC No. ST01849SE, dated 30 January 2012.Flight Structures, Inc. Master Drawing Lists A5FS520-D01, Revision Edted June 15, 2009, or later approved revision, and B0FS015-D01,Revision IR, dated October 20, 2011 or later revision.The Supplemental (...) | Boeing Structural Compliance summary Report Document DLR-36718 revisionNew dated 17 March 2008 must be accomplished concurrently with this STCBoeing Structural Compliance Summary Report Document DLR-36718 RevisionNew dated 17 March 2008 must be acco (...) | 02.02.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|-----------------------|----------------|----------------------|----------------|--|---|--|--|---------------|
| 10016319 | FLIGHT STRUCTURES, Inc. | EASA, I.M.A.S.02627 | REV. 3 | FAA STC ST01950SE | B777-200 | Remove EWIS limitation for EASA STC No. EASA, I.M.A.S.02627 Rev 1 - Door 1 Overhead Flight Crew Rest Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 EWIS The requirements for environmental protection and the associated certificate (...) | FAA STC ST01950SE/SEFSI Master drawing List A7FS700-D01 revision AH dated 5 March 2010/Airplane Flight Manual FSI document TP07-700 AFM revision 1 dated 3 March 2009/FSI Top Collector A700000-001 revision C dated 8 December 2008/B/E Aerospace Drawing 10058 (...) | The Boeing Service Bulletin 777-25-0429 dated 15 October 2008 must be accomplished concurrently with this STC for the serial numbers listed on the FSI top Collector A700000-001 revision C dated 8 December 2008. The Main deck seat arrangement must be as ins (...) | 07.10.2010 |
| 10016564 | FLIGHT STRUCTURES, Inc. | EASA, I.M.A.S.02955 | | FAA STC ST00386SE | BOEING 747-200 | This design change relocates the E/E Bay Acces Hatch on the main deck floor and Installs Gari o Handling System Provisions for a 30pallet main deck cargo configuration per Flight Structures, Inc. | | Master Drawing list 96FS624-D01 at Revision E dated September 18, 2003, or later FAA approved revisions. | This STC can only be installed in aircraft with FAA STC SA2725CE-D or STC ST00255WI-D or STC ST00252WI-D or STC ST00214SE installed. | 18.09.2008 |
| 10016610 | FLIGHT STRUCTURES, Inc. | EASA, I.M.A.S.03030 | REV. 2 | FAA STC ST02004SE | B777-200 | Remove EWIS limitation for EASA STC No. EASA, I.M.A.S.03030 - Door 3 Overhead Attendant Rest Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 EWIS The requirements for environmental protection and the associated certificate (...) | FAA STC ST02004SE/SEFSI Master drawing List A7FS714-D01 revision R dated 5 March 2010/Airplane Flight Manual FSI document TP07-714 AFM dated 16 January 2009/A7FS714-D15 Revision D dated 26 July 2010 Supplemental Instructions for Continued Airworthiness (...) | The Main deck seating arrangement must be installed in accordance with British Airways drawing 3-302-342 revision 10 dated 12 February 2009/FSI STC 10016319 rev. 2 (FAA STC number ST01950SE) for the FSI Door 1 Overhead Flight Crew Rest Installation (OF (...) | 07.10.2010 |
| 10029094 | FLIGHT STRUCTURES, Inc. | P-EASA, I.M.A.S.02978 | REV. 1 | FAA STC ST02051SE | B777-200 | Remove EWIS limitation for EASA STC No. 10029094 - First Class Seat Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 EWIS The requirements for environmental protection and the associated certificate (...) | FAA STC ST02051SE/SEFSI Master drawing List A7FS703-D01 Revision D dated 21 June 2010/A7FS703-D15 Revision C dated 13 July 2010 Supplemental Instructions for Continued Airworthiness or later revisions of the above listed documents approved by EASA in accordance with (...) | Concurrent installation of EASA STC No. 10029095 (EASA Project No. P-EASA, I.M.A.S.03043) is required. Concurrent installation of EASA STC No. EASA, I.M.A.S.01797 is required | 04.10.2010 |
| 10029964 | FLIGHT STRUCTURES, Inc. | P-EASA, I.M.A.S.03045 | REV. 1 | FAA STC ST02060SE | B747-400 | Amendment of STC 10029964 in order to remove EWIS Limitations and update the MDI reference | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 EWIS The requirements for environmental protection and the associated certificate (...) | FAA STC No. ST02060SE dated 23 April 2010/FAA SoC dated 03 May 2010/Flight Structure Inc. Master Drawing List A7FS702-D01 Revision H dated September 22, 2010/The Supplemental Instruction for Continued Airworthiness FSI Document A7FS702-D15 Revision C date (...) | Concurrent Installation of EASA STC No. 10029975 is required. Concurrent installation of FAA STC ST00720LA (EASA approved under grandfather rights by CAA Airworthiness Approval Note No. 26656 dated 8 March 1999) is required | 04.10.2010 |
| 10031815 | FLIGHT STRUCTURES, Inc. | P-EASA, I.M.A.S.03215 | | FAA STC ST02038SE | A330-300 | Installation of an Interior Reconfiguration in A330-300 Series aircraft | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 Enhanced Airworthiness Programme for Aeropl (...) | - FAA STC ST02038SE, dated 24th December 2009- Master Drawing List, ref. doc. No. A8FS827-D01, Revision P, dated 7th July 2010- Airplane Flight Manual Supplement, ref. doc. No. TP08-827AFM, Revision 1, dated 6th January 2010- Supplemental Instruction (...) | Not applicable | 17.09.2010 |
| 10032896 | FLIGHT STRUCTURES, Inc. | | | FAA STC ST02160SE-D | B737-500 | Interior Reconfiguration in accordance with Flight Structures, Inc. (FSI) Master Drawing List B0FS003-D01 Installation of new windscreen, storage bin, storage bin extension kits, moveable under-bin class divider, sidewall panel, lowered ceiling panel and associated installation hardware. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | - Flight structures, Inc. (FSI) Master Drawing List B0FS003-D01 Rev. C dated 01-Dec-2010- Certification Plan Summary B0FS003-D10-1 Rev. C dated 02 December 2010- Supplement to Instructions for Continued Airworthiness (ICA) B0FS003-D15 Rev. B dated 19 Nov (...) | Not Applicable | 07.12.2010 |
| 10033035 | FLIGHT STRUCTURES, Inc. | | | FAA STC ST02161SE-D | B737-300 | Interior Reconfiguration in accordance with Flight Structure, Inc. (FSI) Master Drawing List B0FS003-D02 Installation of new windscreen, storage bin, storage bin extension kits, moveable under-bin class divider, sidewall panel, lowered ceiling panel and associated installation hardware. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | - FSI Master Drawing List B0FS003-D02 Rev. C dated 11-Dec-2010- Certification Plan Summary B0FS003-D10-2 Rev. C dated 13-Dec-2010- Supplement to Instructions for Continued Airworthiness (ICA) B0FS003-D15 Rev. B dated 19 November 2010. FAA "or later rev (...) | None | 15.12.2010 |
| 10036059 | FLIGHT STRUCTURES, Inc. | | | FAA STC SA5738NM | B747-400 | Installation of an extended upper deck crew rest in accordance with Flight Structures Inc. Master Drawing List 90SF004-D01, Revision E, dated July 13, 1992, or later FAA approved revisions. Validation of FAA STC SA5738NM. | The Certification Basis for the original product remains applicable to this certificate/ approval. | None. | None. | 09.08.2011 |
| 10036590 | FLIGHT STRUCTURES, Inc. | | | FAA STC ST00695AT-D | B747-400 | Modification of Upper Deck Crew Rest Compartment in accordance with Timco Master Data List, MDL-01090, Revision E, dated July 22, 1986, or later FAA approved Revision/Validation of FAA STC ST00695AT-D | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | None | None | 21.09.2011 |
| 10037049 | FLIGHT STRUCTURES, Inc. | | | FAA STC ST00840SE | B747-400 | Reconfiguration of the Main Deck of the B747 400 series airplane in accordance with Flight Structures, Inc. Master Drawing List 99FS908-D01 Rev. E, dated 10 November 2009 or a later FAA approved revision. Validation of FAA STC ST00840SE .. | The Certification Basis for the original product remains applicable to this certificate/ approval.,, | Flight Structures, Inc. Master Data List MDL 99FS908-D01 Rev. E. For later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None. | 27.10.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|--------------------|----------------|----------------------|--|---|---|--|---|---------------|
| 10037365 | FLIGHT STRUCTURES, Inc. | | | FAA STC ST00756SE | B747-400 | Reconfiguration of the Main and Upper Deck of B747-400 series airplanes in accordance with Flight Structures, Inc. Master Drawing List 98FS18-D01 Rev. J, dated November 10, 2009 or a later FAA approved revision. Validation of FAA STC ST00756SE. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Structures, Inc. Master Data List MDL 98FS18-D01 Rev. J, dated November 10, 2009; Flight Structures, Inc. Instructions for Continued Airworthiness 98FS18-D08 Rev. G, dated November 5, 2002; or later revisions of the above listed documents approved (...) | None. | 22.11.2011 |
| 10039031 | FLIGHT STRUCTURES, Inc. | | | FAA STC ST02164SE | A330-301/-302/-303 A330-321/-322/-323 A330-341/-342/-343 | A330-300 Cabin reconfiguration. This STC installs new economy class and business class seats together with the replacement of the IFE system. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. EASA TCDS A.004 Special Condition E-1014, Issue 4: HIC Compliance (Inflatable Restraints (...)) | Master Drawing List, Ref: B0FS014-D01, Revision E, dated 23 March 2012. Instruction for Continued Airworthiness Supplement, Ref: B0FS014-D15, Revision A, dated 12 March 2012. Aircraft Flight Manual Supplement, Ref: TP10-014AFM, Revision 1, dated 23 March (...) | None | 30.03.2012 |
| 10041241 | FLIGHT STRUCTURES, Inc. | | | FAA STC ST01941SE | A300 B4-601/-603/-605R A300 B4-620/-622/-622R | Validation of FAA Supplemental Type Certificate (STC) ST01941SE Conversion of an Airbus A300B4-600 series aeroplanes from a passenger configuration to a freighter configuration (including main deck cargo door, interior, and associated systems), in accordance with (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The certification basis is CS25 original issue for Significant Changed Areas and in accordance with (...) | B/E Aerospace A300-600 Aeroplane Flight Crew Operating Manual Supplement Document TP1-106FCOAAAR Instructions For Continued Airworthiness A300-600 FSI MDCLS P/N58535-101 No. 2148 original issue, dated 16 July 2007. Aeroplanes converted by this STC must (...) | The STC is limited to three A300B4-600 series model aeroplanes, Manufacturer Serial Number (MSN) 734, 739 and 756. Per the Temporary Deviation, aeroplanes converted under this STC, are limited to 1,500 flight cycles or 3,300 flight hours, whichever occurs (...) | 31.08.2012 |
| 10042096 | FLIGHT STRUCTURES, Inc. | | | FAA STC ST02166SE | 747-400 | Installation of additional Crew Rest Bunks in the Door 5 Overhead Crew Rest. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The Certification basis for this modification is in accordance with Type Certificate Da (...) | Validation of FAA STC ST02166SE Cert Plan B1FS103-D10 d.d 23-9-2012 Master Drawing List B1FS103-D01 Revision B d.d. 26-10-2012 The Airplane Flight Manual Supplement, FSI document TP11-103AFM d.d. 24-10-2011 The supplemental Instructions for Continued Air (...) | Maximum number of occupants of the Door 5 overhead crew rest is eight persons. | 08.11.2012 |
| 10043740 | FLIGHT STRUCTURES, Inc. | | | FAA STC ST01493SE | 777-200/200LR | Interior Reconfiguration (description: Installation of business class seats <-(-> new RH & LH closets, fold down tables, sidewall up wash and ceiling downwash, full spectrum, digital LED lighting system, new economy class quintuple seats & economy class (...)) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List, Flight Structures Document A2FS237-D01, Revision T, dated July 19, 2010; Supplemental Instructions for Continued Airworthiness, Flight Structures Document A2FS237-D15, Revision A, dated January 6, 2005; Airplane Flight Manual Sup (...) | EASA validated version of FAA STC ST01494SE for installation of lavatory and STC ST02044NY for installation of an inflight entertainment system must be installed concurrently with this STC. Prior to installation of this design change it must be determined (...) | 21.02.2013 |
| 10045816 | FLIGHT STRUCTURES, Inc. | | | FAA STC ST02169SE | 747-400 | Interior reconfiguration of KLM B747-400 Aircraft Installation of Business Class Ottomans, a new Closet, new Class Divider, modification of Pilot Crew Rest, new Storage Bin and new Upper Deck Decompression Venting. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: 14 CFR Part 25 at amendment 25-54; 25.365(e)(2), (g); 25.571 (a)(b); EASA CS25 Amendment level 12; 25.365(a), (b), (c), (d); 25.61325 (...) | B2FS202-D01, Revision J, 10 July 2013 - Master Drawing List B2FS202-D10, Revision B, 11 July 2013 - Certification Plan B2FS202-D15, Revision IR, 09 May 2013 - Instructions for Continued Airworthiness B2FS202-ST-01, 11 July 2013 - Applicant Statement of C (...) | This STC is a prerequisite of the KLM STC with project number 010018346-001. The following STCs should be installed simultaneously: STC EASA.I.M.A.S.03182, Delta Engineering - Update of Panasonic IFE/ISPS configuration (amendment of previous approved (...)) | 19.07.2013 |
| 10045998 | FLIGHT STRUCTURES, Inc. | | | FAA STC ST02156SE-D | A330-200 A330-300 | Installation of an interior Reconfiguration, ZONE A - FIRST AND BUSINESS CLASS Installation of a new underbin closet, new VCC and new F/C and B/C seats ZONE B - TOURIST CLASS Installation of new T/C seats, outboard closets forward of Door 2L and 2R, rear (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification Plan A8FS804-D10-EASA, rev. A; MASTER DRAWING LIST, ref A8FS804-D01, rev. AC, ICA A8FS804-D15, rev. D; AFMS Supplement, TP08-804AFM, rev. 2; FAA Letter 150S-13-226, dated on 26th July 2013 or later revisions of the above listed documents approved (...) | 1) Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 05.08.2013 |
| 10046398 | FLIGHT STRUCTURES, Inc. | | | FAA STC ST01492SE | 747-400 | Interior Reconfiguration Installation of first class seats and ottomans, minipod business class seats, VIP lavatories, sidewall credenzas, full spectrum light system, electric window shades, In Seat Power Supply System, Video Control Center (VCC), over (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | A2FS238-D01, Revision T, 2013 August 16 - Master Drawing List A2FS238-D10, Revision C, 2009 April 18 - Certification Plan A2FS238-D15, Revision IR, 2005 February 21 - Instructions for Continued Airworthiness TP2-238AFM, 2004 December 20 - Airplane Flight Manual Supplement, ref B1FS115-D15, (...) | The following STCs must be installed concurrently with this STC: EASA STC 10042094 (FAA STC ST01514SE) - Lavatory and galley modification; EASA STC 10042191 (FAA STC ST02049NY) - Inflight Entertainment System. The following FAA approved B/E Aerospace SB (...) | 16.09.2013 |
| 10046481 | FLIGHT STRUCTURES, Inc. | | | FAA STC ST02171SE | A340-600 | Installation of Cabin Modification Provisions. Install new and modify existing electrical, structural and systems provisions only for the installation of a new cabin interior. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification Plan B1FS115-D10, rev. A; Master Drawing List, ref B1FS115-D01, rev. N; ICA : Supplement To Instructions For Continued Airworthiness For The Installation Of Cabin Modification Provisions On Lufthansa A340-642 Airplanes, ref B1FS115-D15, (...) | 1) Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 25.09.2013 |
| 10046520 | FLIGHT STRUCTURES, Inc. | | | ST02170SE | A340-600 | Lower Lobe Fixed Crew Rest Installation. The installation is comprised of a new lower deck crew rest, cargo loading equipment and installation kit. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): D-01 : Lower Deck Crew Rest Compartment Installation. The requirements for environmental protection and (...) | Certification Plan B1FS102-D10, rev. C; Master Drawing List, ref B1FS102-D01, rev. P; ICA : Instructions For Continued Airworthiness, ref B1FS102-D15, rev. B; AFM Supplement, TP11-102AFM, rev. 2; FAA Letter 150S-13-288, dated on 26th September 2013 (...) | 1) Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 27.09.2013 |
| 10015410 | FLIGHT TEST ASSOCIATES, INC | EASA.I.M.A.S.01124 | | FAA STC ST01524LA | LEARJET 31 LEARJET 35 LEARJET 35A LEARJET 36 LEARJET 36A | Installation of Rosemount Pitot Static Probe Part Numbers 856NA1 & 856NA2 | | West Star Aviation Inc. Master Drawing List 30A210001 Revision G dated May 20 2004 or later approved revision, for the airplanes equipped with J.E.T. Model FC-200 Autopilots. AFM Supplement 30A040002 Revision B dated August 10 2004 or later approved r (...) | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 05.07.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|---------------------|----------------|------------------------------|---|--|--|---|--|---------------|
| 10015411 | FLIGHT TEST ASSOCIATES, INC | EASA.IM.A.S.01125 | | FAA STC ST01525LA | LEARJET 31 LEARJET 35 LEARJET 35A LEARJET 36 LEARJET 36A | Installation of Honeywell AZ-252 air data computer, Ba-250 barometric altitude display, AM-250 barometric altimeter, AL-800 altitude alert controller, and Aerosemic standby altimeter. | | West Star Aviation Inc. Master Drawing List 30A210002 revision F dated July 12 2004 or later approved revision. For airplanes equipped with J.E.T. model FC-200 autopilots Flight Test Associates AFM supplement 30A040002 revision B dated August 12 2004 (...) | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 06.07.2005 |
| 10042951 | FLIGHT TEST ASSOCIATES, INC | EASA.IM.A.S.01126 | | FAA STC ST0152LA & ST01525LA | LEARJET 35, 35A, 36, 36A | Aircraft modifications for establishing the approval basis for Reduced Vertical Separation Minimum. | | FAA STC ST0152LA & ST01525LA dated August 12 2004 or later approved revisions. Learjet models 35/35A/36/36A with J.E.T. models FC 200 & FC 350 autopilots AFM Supplement 30A040003 revision B dated August 10 2004 or later approved revision. Instructions to (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 07.06.2005 |
| 10015560 | FLINT AERO, INC. | EASA.IM.A.S.01457 | | FAA STC SA3232NM | CESSNA 206H CESSNA T206H CESSNA TU206G CESSNA U206G | Installation of 14.9 gallon usable fuel containing lightning resistant wing tips | | Appendix to AFMS for Soly Turbine converted Cessna Model Uand TU 206G and 206H/T206H Airplanes for Flint Aero, Inc. Extended Wing Tip Fuel Tanks Flint Aero STC SA3232NM and Soly STC SA2353NM | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. Th (...) | 23.02.2009 |
| 10016592 | FLINT AERO, INC. | EASA.IM.A.S.03003 | | | CESSNA 182 CESSNA 182A CESSNA 182B CESSNA 182C CESSNA 182D CESSNA 182E CESSNA 182F CESSNA 182G CESSNA 182J CESSNA 182K CESSNA 182M CESSNA 182N CESSNA 182P CESSNA 182Q CESSNA 182R CESSNA 182S CESSNA 182T CESSNA R182 CESSNA T182, T182T CESSNA TR182 | Auxiliary Internal Wing Fuel Tanks. This STC is a revision of LBA STC SA 1305, dated 14 January 2003. | | None. | None. | 15.08.2008 |
| 10031241 | FLINT AERO, INC. | P-EASA.IM.A.S.02712 | | FAA STC SA3226NM | 210G, 210H, 210J, 210K, 210L 210M, 210N T210G, T210H, T210J, T210K T210L, T210M, T210N, P210N | Installation of 16,25 US Gallon usable fuel lightning resistant wing tip fuel tanks | The Certification Basis for the original product remains applicable to this certificate/ approval. CAR 3 The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and (...) | Installed in accordance with Flint Aero Master Drawing List MDL FALS210N/CFAA approved Flint Aero AFM-S dated 16 Jan 1984 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | According limitation section of AFM-S for P210N and T210N following limitation additionally apply:- Wing Tip Aux/ Aux Tank either is less than 7 gals MTOW 3730lbs, MLW 3530lbs-Wing Tip Aux/ Aux Tank both are 7 gals or more MTOW 4000lbs, MLW 3800lbs | 03.08.2010 |
| 10034927 | FLINT AERO, INC. | | | FAA STC SA2529WE | 185, 185A, 185B, 185C, 185D 185E, A185E, A185F | Installation of twelve (12) gallon auxiliary fuel tank in each wing inacc. with FAA STC SA2529WE. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Instruction Flint Aero, Inc. Doc. No. FA180, Rev. L, dated September 4, 2007 Operational Limitation: see placards in acc. with Flint Aero, Inc. Doc.No. FA180, Rev. L, dated September 4, 2007 Maintenance Instruction in acc. with Flint Aero, (...) | The limitations and conditions of FAA TCDS 3A24 apply, except as outlined below: a) Model 185 through A185F S/N 18502262 - 89 gal. (85 gal. useable); two 32.5 gal. main tanks in wings at (+48) and two 12 gal. (11.5 gal. useable) aux. tanks in wings at (+ (...) | 13.05.2011 |
| 10035079 | FLINT AERO, INC. | | | FAA STC SA1614WE | SEE EASA APPROVED MODEL LIST | Installation of one twelve (12) gallon auxiliary fuel tank in each outboard wing panel in acc. with FAA STC SA1614WE. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Flint Aero Master Drawing List 15171821, Rev. A, FAA approved November 5, 2007 or Sealed Flint Aero Master Drawing List FA170, last Rev. Installation instructions FA180, Rev. L, dated September 4, 2007 Airplane Flight Manual Supplement FT453.001, IR, FAA (...) | The limitations and conditions of FAA Aircraft Specification A-799, FAA Data Sheets 3A12, 3A17, or FAA Data Sheets A4EU and A18EU or EASA TCDS IM.A.051 apply, except as outlined below of this STC: 1. Fuel Capacity Add *24 gal. (23 gal. useable); two 12 (...) | 24.05.2011 |
| 10014390 | FLYBE LIMITED | EASA.A.S.01676 | | | BAE ATP | Cargo Layout for class "E" compartment (LFD) | | FlyBe Modification DEC-E1000-FlyBe ATPF Flight Manual Supplement Number ATP 1 LFD, Issue 1 | | 30.01.2006 |
| 10014413 | FLYBE LIMITED | EASA.A.S.01712 | | | BAE ATP | Introduction of Class E Compartment Cargo Layout | | FlyBe Modification Statement BE/MOD/E750FlyBe AFM Supplement ATP-AFM-01 Aircraft Loading Manual Supplement ATP-LM-01 Engineering Manual Supplement ATP-EMS-1 | Restricted to Aircraft Serial Number 2042 & 2052 | 17.01.2006 |
| 10042037 | FLYBE LIMITED | | REV. 1 | | ERJ 190-100 STD/LR/IGW/ECJ/SR ERJ 190-200 STD/LR/IGW | Embraer ERJ 190 - Installation of Tropospheric Airborne Data Reporting System This EASA approval is an update of the EASA approval ref. 10042037 Description as per Associated Technical Documentation | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List ref. RE2088-06 Issue 3, dated 07/11/2012 AFM Supplement RE2088-18 Issue 2, dated 06/11/2012. ICA Supplement RE2088-17 Issue 3, dated 01/10/2012 or later revisions of the above listed documents approved by EASA | ALS update as stated in ICA Supplement RE2088-17 Issue 3, dated 01/10/2012. | 08.11.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|-------------------------|---|---|---|---|--|---------------|
| 10016123 | FLYHT AEROSPACE SOLUTIONS Ltd. | EASA.IM.A.S.02316 | | TCCA STC SA06-33 | AIRBUS A319 AIRBUS A320 AIRBUS A321 | AFIRS 220 Automated Flight Information Reporting System Installation with optional Global Voice and Global Text Satellite Communication/EASA Validation of Transport Canada approved Supplemental Type Certificate SA06-33. | | ∫ TCCA STC SA06-33∫ AMS Master Document List, Ref. 121-0029, Rev. -- dated November 24th 2006∫ AMS Installation Instructions, Ref. 151-0021, Rev. A, dated November 23rd 2006 | ∫ AMS Instructions for Continued Airworthiness, Ref. 153-0029, Rev. A. Compliance with the approved Airworthiness Limitations contained therein is mandatory. | 20.07.2007 |
| 10016213 | FLYHT AEROSPACE SOLUTIONS Ltd. | EASA.IM.A.S.02461 | | TCCA STC SA07-47 | BOEING 757-200 | AFIRS 220 Automated Flight Information Reporting System Installation with Optional Global Voice and Global Text Satellite Communication. | | AMS Installation Instructions, 151-0024, Rev. --, 27/03/2007. Instructions for Continued Airworthiness, 153-0030, Rev. --, 01/12/2006. | | 13.09.2007 |
| 10016343 | FLYHT AEROSPACE SOLUTIONS Ltd. | EASA.IM.A.S.02659 | | TCCA STC SA07-7 | BOMBARDIER DHC-8-100 SERIES BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-300 SERIES | Aeromechanical Services AFIRS (Automated Flight Information Reporting System) 220 installation with optional Global Voice and Global Text Satellite Communication. | | AMS Master Document List, 121-0036, Rev. A, 1. November 2007/AMS Installation Instructions, 151-0023, Rev. B, 1. December 2006/AMS Instructions for Continued Airworthiness, 153-0031, Rev. A, 1. December 2006 | | 21.01.2008 |
| 10016344 | FLYHT AEROSPACE SOLUTIONS Ltd. | EASA.IM.A.S.02660 | REV. 2 | TCCA STC SA07-8 | BOEING 737-200 SERIES BOEING 737-300 SERIES BOEING 737-400 SERIES BOEING 737-500 SERIES BOEING 737-600 SERIES BOEING 737-700 SERIES BOEING 737-800 SERIES | AFIRS 220 Automated Flight Information Reporting System installation with Optional Global Voice and Global Text Satellite Communication. | | AFMS, Wiring Diagram Manual or later revisions of this document approved by EASA in accordance with EASA ED Decision 2004102CF (or subsequent revisions of this decision). | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 22.05.2008 |
| 10016385 | FLYHT AEROSPACE SOLUTIONS Ltd. | EASA.IM.A.S.02711 | | TCCA STC SA07-84 | BOMBARDIER CL-600-2B19 (SERIES) | AFIRS 220 Automated Flight Information Reporting System installation with Optional Global Voice and Global Text Satellite Communication | | TCCA Statement of Compliance ref. C-07-0944 dated 02 July 2008/AMS Master Document List, Ref. 121-0032 Rev. - dated 29 June 2007 or later approved revision/AMS Installation Instructions, Ref. 151-0027 Rev. - dated 27 March 2007 or later approved revision | AMS Instructions for Continued Airworthiness, Ref. 153-0036 Rev. A or later approved revision. Compliance with the approved Airworthiness Limitations contained therein is mandatory. | 03.07.2008 |
| 10016522 | FLYHT AEROSPACE SOLUTIONS Ltd. | EASA.IM.A.S.02895 | | TCCA STC SA07-124 | BOEING 767-200 BOEING 767-300 | AFIRS 220 Automated Flight Information Reporting System installation with Optional Global Voice and Global Text Satellite Communication | | AMS Master Document List, Ref. 121-0039 Rev. B, Dated 3 July 2008/AMS Installation Instructions, Ref. 151-0030 Rev. A, Dated 12 November 2007 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The i (...) | 19.08.2008 |
| 10027904 | FLYHT AEROSPACE SOLUTIONS Ltd. | EASA.IM.A.S.03184 | REV. 4 | TCCA STC SA08-7 | HAWKER 750 HAWKER 800XP HAWKER 850XP HAWKER 900XP | AFIRS 220 / AFIRS 228 Automated Flight Information Reporting System Installation With Optional Global Voice and Global Text Satellite Communications. | AFIRS 220 System installation is FAR 25 including Amendments 25-1 through 25-97. AFIRS 228 System installation is FAR 25 including Amendments 25-1 through 25-97 and FAR 25.981 at Amendment 25-102. | AFIRS 220 System Installation:- Installation of AFIRS 220 system must be in accordance with AeroMechanical Services Ltd ("FLYHT"), Installation Instructions 151-0032, Rev. E, Transport Canada (TCCA) approved 18 October 2010, or later TCCA approved revision (...) | None. | 29.08.2012 |
| 10028338 | FLYING COLOURS CORP. | | REV. 2 | TCCA LSTC C-LSA09-326/D | CL-600-2B19 | Installation, Complete Custom Aircraft Interior | EASA.IM.A.023The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product areunch (...) | TCCA LSTC No. C-LSA09-326/D Issue 1 dated 24 November 2009. Aero Consulting Services Ltd. Modification Data Summary G288000 Revision G. Flying Colours Corp. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior G288001 Revision NC (...) | Restricted to ASN 8092 | 14.12.2011 |
| 10032278 | FLYING COLOURS CORP. | | REV. 1 | TCCA STC NO. SA10-66 | CL-600-2B19 | Installation, Equipment Bay Catwalk | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product areunch (...) | TCCA SoC ref. C-10-0781 dated 07 October 2010. TCCA STC No. SA10-66 Issue 1 dated 31 August 2010. ACS-NAI Ltd Modification Data Summary, G744000 Revision A. Flying Colours Corp. Equipment Bay Catwalk Instructions for Continued Airworthiness, G744071 Rev (...) | none | 28.10.2010 |
| 10032279 | FLYING COLOURS CORP. | | REV. 1 | TCCA STC NO. SA10-67 | CL-600-2B19 | Installation, Equipment Bay Ladder | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product areunch (...) | TCCA SoC ref. C-10-0780 dated 07 October 2010. TCCA STC No. SA10-67 Issue 1 dated 31 August 2010. ACS-NAI Ltd. Modifications Data Summary G745000 Revision A. Flying Colours Corp. Equipment Bay Ladder Stowage Provisions Instructions for Continued Airworthiness (...) | Prior embodiment of EASA STC No. 10032278 'Installation, Equipment Bay Catwalk' is required. | 28.10.2010 |
| 10032280 | FLYING COLOURS CORP. | | REV. 1 | TCCA STC NO. SA10-65 | CL-600-2B19 | Installation, Equipment Bay Access Door Steps | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product areunch (...) | TCCA SoC ref. C-10-0779 dated 07 October 2010. TCCA STC No. SA10-65 Issue 1 dated 01 September 2010. ACS-NAI Ltd. Modification Data Summary, G743000 Revision A. Flying Colours Corp. Equipment Bay Access Door Steps Instructions for Continued Airworthiness (...) | none | 28.10.2010 |
| 10039852 | FLYING COLOURS CORP. | | | TCCA STC NO. SA12-33 | CL-600-2B19 | Installation, Aft Equipment Bay Tow Bar Stowage Provisions | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product areunch (...) | Transport Canada STC No. SA12-33 Issue 1 dated 11 April 2012. ACS-NAI Ltd. Modification Data Summary H477000 Revision NC. Flying Colours Corp. Instructions for Continued Airworthiness for the Installation, Tow Bar Stowage Provisions, H477071-ICA Revisi (...) | none | 30.05.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|----------------------|--|---|--|--|---|---------------|
| 10045398 | FLYING COLOURS CORP. | | | C-LSA12-029/D | CL-600-2B19 (CRJ-200) | Corporate Shuttle Conversion | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. C-LSA12-029/D Issue 1, dated 22 August 2012. ACS-NAI Ltd. Modification Data Summary H366000 Revision NC.Flying Colours Corp Airplane Flight Manual Supplement for Corporate Shuttle Conversion, H366000 Revision NC.ACS-NAI Ltd. (...) | Restricted to ASN 7478, 7494, 7625, 7629, 7862, 7864. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce (...) | 21.06.2013 |
| 10016455 | FLYING INVESTMENTS, LLC | EASA.IM.A.S.02801 | | FAA STC ST01349WI | FAN JET FAN JET FALCON SERIES C FAN JET FALCON SERIES D FAN JET FALCON SERIES E FAN JET FALCON SERIES F FAN JET FALCON SERIES G MYSTERE FALCON 20-C5 MYSTERE FALCON 20-D5 MYSTERE FALCON 20-E5 MYSTERE FALCON 20-F5 | Reduced Vertical Separation Minimum (RVSM) group airworthiness approval of specific Dassault Aviation Fan Jet Falcon and Mystere-Falcon 20 aircraft in accordance with Flying Investments LLC Master Drawing List MDL1017, Revision G, dated 19th February 2008 or later EASA approved Revisions. | | Flying Investments LLC Master Drawing List MDL1017, Revision G, dated 19th February 2008. Applicable Flight Manual Supplements: # Flying Investments LLC AFM Supplement KSR-0302202/Revision C dated 24th January 2008 for Fan Jet Falcon, Fan Jet Falcon (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 22.07.2008 |
| 10016456 | FLYING INVESTMENTS, LLC | EASA.IM.A.S.02802 | | FAA STC ST00571DE | FAN JET FAN JET FALCON SERIES C FAN JET FALCON SERIES D FAN JET FALCON SERIES E FAN JET FALCON SERIES F FAN JET FALCON SERIES G MYSTERE FALCON 20-C5 MYSTERE FALCON 20-D5 MYSTERE FALCON 20-E5 MYSTERE FALCON 20-F5 | Installation of an Innovative Solutions & Supports (IS&S) Dual Air Data Display Units (ADDU), an Analog Interface Unit or Air Data Sensor and Goodrich Dual Pitot-Static Probes, in accordance with Flying Investments Master Data List GEN-031217-001, Re (...) | | Master Data List GEN-031217-001, Revision K, dated 04th December 2007. Applicable Flight Manual Supplements: # Flying Investments LLC AFM Supplement GEN-031217-038 Revision C dated 28th September 2007 for Fan Jet Falcon, Fan Jet Falcon Series C, D, (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 22.07.2008 |
| 10041003 | FMD FLUGDIENST MAGDEBURG GmbH | EASA.A.S.03983 | | | 1. 17ZE, 17ZF, 17ZH, 17ZK, 17ZL 1. 17ZM, 17ZN, 17ZP, 17ZR, 2. FR17ZE | | Installation of vaccine dropping device (SURVIS) (Fuchs Koederabwurfanlage). | Pilot's Operating Handbook and Flight Manual Supplement for Cessna 172, issued 15 July 2009. | The certificate shall remain valid unless otherwise surrendered or revoked. | 29.07.2009 |
| 10014607 | FOKKER SERVICES B.V. | EASA.A.S.02732 | REV. 2 | | CL-600-2C10 (CRJ-700) | VIP conversion of Bombardier CL600-2C10 (CRJ-700) for Consolidated Contractors Company, as detailed in Fokker Services Engineering Report FS-N594, Description of the modification. Revision 2: Wireless handsets in the Aircraft telephone installation in a (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI F-1 "Installation of wireless VOIP telephone handsets" (Means of Compliance) The cer (...) | AFM Supplement n07-FS1-1: Airplane Flight Manual Supplement Bombardier CL600-2C10 Aircraft, CRR-0018, issue 1, certification of provision for wireless telephone communication system dated 20/01/2010. | If the utilisation of the VIP converted aircraft deviates from the assumed utilisation (e.g. flight hours or flight profile) supporting the scheduled maintenance requirements as defined by Bombardier, the operator of the aircraft will have to review the (...) | 21.01.2010 |
| 10014752 | FOKKER SERVICES B.V. | EASA.A.S.02937 | REV. 2 | | A319CJ | Changes to Airbus A319CJ in addition to the VIP Conversion introduced by EASA STC A.S.02937 as described in Fokker Services B.V. Engineering Report A319-SAG-CRR-0002 Issue 1. This STC has been revised to clarify the text in the Limitations and Conditions. | The Certification Basis for the original product remains applicable to this certificate/ approval. | AFM Supplement A319CJ 0.01.00 for STC P. EASA.A.S.02937 or later revisions of the above listed documents approved by EASA | The activation of the Zonal Dryer/Humidification System is not part of this Major Change. The installation of the Zonal Dryer/Humidification System is certified, however the correct use and functioning is not certified. Therefore the Zonal Dryer/Humidification System may not be deactivated. | 01.07.2010 |
| 10031332 | FOKKER SERVICES B.V. | EASA.A.S.03317 | | | A318-112 | A318-112 VIP conversion, for description see Chapter 1.1 of Compliance Record Report A18-MAK-CRR-0001 issue 5 d.d. 29-5-2009 | | -, Compliance Record Report A18-MAK-CRR-0001 issue 5 d.d. 29-5-2009, -AFM Supplement Airbus A318 Model A318-112 Aircraft for the VIP Conversion As installed in Serial number 3238 issue 5 d.d. 29-5-2009, -MMEL Supplement Airbus A318 Model A318-112 Air (...) | 1. Refer to Chapter 2, Limitations of the AFM Supplement Airbus A318 Model A318-112 Aircraft for the VIP Conversion As installed in Serial number 3238 issue 5 d.d. 29-5-2009. The STC is approved only for the product configuration as defined in the appr (...) | 29.05.2009 |
| 10034242 | FOKKER SERVICES B.V. | P-EASA.A.S.03722 | | | AIRBUS A320-232 | Airbus A320-232 VIP Conversion. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. JAR 25 Change 11C/RI C-01 Large Display Monitors CRI D-01 Isolated Compartments CRI D-02 (...) | A320-CAI-CRR-0001 issue 3, approval date 17th March 2011. A320-CAI-AFM-S-APPROVAL issue 1, dated 15th March 2011. A320-CAI-MMEL-S-APPROVAL issue 1, dated 15th March 2011. CA Supplement document: SE-712 issue 2, dated 18th February 2011 or later revisions of the above listed documents approved by EASA. | Refer to Limitations section of the approved AFM Supplement A320-CAI-AFM-S-APPROVAL issue 1, dated 15th March 2011. Remarks: The aircraft type, model and manufacturer serial number for which this change is applicable is: Airbus A320-232 with Manufacturer/Serial Number 1868. | 18.03.2011 |
| 10038053 | FOKKER SERVICES B.V. | | | | F28 MARK 0070 | Activation of WLAN functionality of an already installed and certificated router (ref EASA STC 10027637) The Fokker 70 aircraft contains a router that has been certificated under STAC 10027637. Fokker Services will activate and certify the WLAN functionality. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Acceptable Means of Compliance: CRI F-21 "Wireless Local Area Network (WLAN)" The requirem (...) | -CRI F-21 "Wireless Local Area Network (WLAN)" -Certification Plan CP-0119 Issue 2-AFM change AFM/JET2011-007 Issue 1-Compliance Record Report CRR-0119 Issue 1 for later revisions of the above listed documents approved by EASA | -Installation of EASA STC 10027637 is a prerequisite for this STC-Limitation of Wireless Local Area Network as documented in AFM change AFM/JET2011-007 | 16.01.2012 |
| 10042160 | FOKKER SERVICES B.V. | | | | B737-600/-700/-800/-900/-900ER | Introduction of LED light assemblies in the Boeing B737 NG aircraft | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification Plan CP-0101, Issue 1 Certification Record Report CRR-0101, Issue 1 CA Supplement SE-750r later revisions of the above listed documents approved by EASA | none | 12.11.2012 |
| 10048185 | FOKKER SERVICES B.V. | | | | DHC-8-102, DHC-8-103 DHC-8-106 DHC-8-201, DHC-8-202 DHC-8-301, DHC-8-311 DHC-8-314, DHC-8-315 DHC-8-401, DHC-8-402 | Provisions for a Class 2 Electronic Flight Bag (EFB) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification Reports CRR-0113, CRR-0138, CRR-0179 or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 26.08.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|---------------------|----------------|----------------------------------|--|---|--|---|---|---------------|
| 10046397 | FOKKER SERVICES B.V. | | | | A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-111, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231, A321-232 | Introduction of LEDs as a replacement of the existing Passenger Compartments Tube Lights. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification Plan CP-0156 Issue 1 Compliance Record Report CRR-0156 Issue 1 Instructions for Continued Airworthiness report SE-760 Issue 1 for later revisions of the above listed documents approved by EASA. | This STC is applicable to MSN 3151 and below only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no (...) | 16.09.2013 |
| 10015713 | FOKKER50FREIGHTER STC CORP. | EASA, I.M.A.S.01710 | REV 1 | FAA STC ST01916LA | FOKKER F.27 MARK 050 | Installation of Large Cargo door. Installation of Large Cargo door. | | 1. Maintenance Manual Supplement nr. MMS 222 Rev. B.2. FAA approved Airplane Flight Manual Supplement document nr. FMS 223 dated October 30, 2006.3. FAA approved Fokker 50 Freighter STC Corporation Master Drawing List 24B500 revision NC dated September 14, 2006.1. Maintenance Manual Supp. | 1. To allow the carriage of cargo STC I.M.A.S.01711 shall be incorporated.2. The maximum number of occupants in the aircraft including crew members is 3:3. This STC is approved only for the product configuration as defined in the approved des (...) | 21.05.2008 |
| 10015714 | FOKKER50FREIGHTER STC CORP. | EASA, I.M.A.S.01710 | REV 1 | FAA STC STO1916LA - EASA, I.M.A. | FOKKER F.27 MARK 050 | | | | | 21.05.2008 |
| 10015715 | FOKKER50FREIGHTER STC CORP. | EASA, I.M.A.S.01711 | REV 1 | FAA STC ST00260DE | FOKKER F.27 MARK 050 | Installation of Class E Cargo Interior, configuration 24B301-211 in accordance with Branson Aircraft Corporation Drawing List 24A300, revision B dated November 5, 1996, FAA approved November 7, 1996 or later revision approved in accordance with ED De (...) | | 1. Cargo Loading Manual WB-2052 revision or later revision approved in accordance with ED Decision 2004/04/CF dated 10 December 2004.2. FAA approved Airplane Flight Manual Supplement nr. FMS 232, dated October 30, 2006 or later revision approved in a (...) | 1. Cargo Loading Manual WB-2052 Rev. C or later approved revisions shall be used when loading the aircraft.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" a (...) | 02.11.2007 |
| 10028703 | FORCED AEROMOTIVE TECHNOLOGIES | | | FAA STC SE10926SC | IO-550-N | Installation of BAR Air, LLC supercharger system on Teledyne Continental Motors IO-550-N Engines in accordance with Master Drawing List MDL-SR22-1-E, Revision A, dated April 4, 2008 or later FAA approved revision - EASA Validation of FAA STC SE10926SC. | JAR-E Change 9 dated May 4, 1990. The Certification Basis for the original product remains applicable to this certificate except for JAR-E 10, E 20, E 25, E 40, E 70, E 220, E 270, E 360 which are amended by JAR-E Change 10 | Instructions for Continued Airworthiness are contained in Document No.22007-30, Revision N/C, dated December 1, 2007 or later revisions approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | The supercharged IO-550-N engine is limited to 29.6 in Hg (204 kPa) MAP and 2700 RPM or 310 HP (231 kW). | 03.02.2010 |
| 10029290 | FORCED AEROMOTIVE TECHNOLOGIES | | | FAA STC SA10925SC | SR22 | Installation of BAR Air, LLC modified Teledyne Continental Motors IO-550-N law. MDL-SR22-1-A, Rev. E or later FAA approved revisions. Installation requires EASA STC 10028703 to be installed. | CS-23.B The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged (...) | MDL-SR22-1-A, Rev E Airplane Flight Manual Supplement 22006, Rev AICA 22006-30, Rev E or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | As referred in limitations section of the AFM-S 22006 Engine must be modified acc. EASA STC 10028703 | 17.03.2010 |
| 10029878 | FORCED AEROMOTIVE TECHNOLOGIES | | | FAA STC SE10233SC | O-470-L, -R, -S, -U | Installation of a Belt-Driven Vortech V-15 supercharger assembly on the Teledyne Continental Motors (TCM) O-470-L, O-470-R, O-470-S, and O-470-U engines in accordance with Type Inspection Data contained in Master Drawing List No. MDL-182-1-E, Rev. D, da (...) | CAR 13 The Certification Basis for the original product remains applicable to this certificate, except for JAR-E 10, E 20, E 25, E 40, E100(b), E80, E70, E 200, E 270, E 360 which are amended by JAR-E Change 10 | Instructions for Continued Airworthiness are contained in Document No.22002-05, Rev. A, dated January 4, 2005, or later revisions approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | The supercharged O-470-L, -R, -S, and -U engines are limited to 28 inHg (193 kPa) MAP and 2600 RPM or 230 HP (171.5 kW) with minimum fuel grade Avgas 100LL. | 28.04.2010 |
| 10030005 | FORCED AEROMOTIVE TECHNOLOGIES | | | FAA STC SE10233SC | 182E, 182F, 182G, 182H, 182J, 182K, 182L, 182M, 182N, 182P, 182Q, 182R | Installation of Rod Sage Supercharged Teledyne Continental Motors (TCM) O-470-L, O-470-R, O-470-S and O-470-U Engines inacc. with FAA STCSA10232SC | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Drawing List MDL-182-1-A, Rev. F, dated January 4, 2005. Supercharger Installation Instructions, Drwg. No. II-182, Rev. F, dated January 4, 2005. Supercharger Installation Instructions, Drwg. No. EI-182, Rev. B, dated January 4, 2005. Instruction for (...) | This Supplemental Type Certificate requires installation of EASA STC 10029878 (see also FAA-STC SE10233SC) | 10.05.2010 |
| 10033805 | FR AVIATION Ltd. | | | | B300 | Observer Training Platform Mod.-B300. | CS-23.B | Installation must be in accordance with Master Document List included in the Design Change Statement DC-0166 Iss. 1 dated 21st January 2011 or later Approved Revisions. Operation must be in accordance with Pilots Operating Handbook Supplement No. FRAH/FM/0 (...) | The approval is limited to the Aircraft Serial Numbers: FL-597<->-<->-FL-618, FL-629 and FL-633. | 09.02.2011 |
| 10033806 | FR AVIATION Ltd. | EASA, I.M.R.S.01558 | | FAA STC SH2735NM | MD 369D, 369E, 369H, 369HE MD 369HS, 500N, 600N | Pneumatic Door Opener (PDO) This EASA STC supersedes and is a revision of LBA STC 0426/9036. | CS-27 | -Master Drawing List: DL-369PDO-100 Rev. E or later approved revision-Instructions for Continued Airworthiness: PR-369DA-120M Rev. O or later approved revision-Summary Document (Including Cert. Basis): Parawon letter to EASA of 5th August 2008 | None | 23.06.2009 |
| 10035091 | FR AVIATION Ltd. | | | | TYPE: FAN JET FALCON BASIC, D AND E SERIES | Extended underwing stores carriage (flutter clearance) The goal is to approve the flutter clearance of underwing store configurations installed onto Fan Jet Falcon aircraft with existing underwing hard points. This design definition establishes an enve (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. | Design Change Statement doc. N° DC-0194 Issue 1 dated 20.05.2011. Compliance Record doc. N° CR-0194 Issue A dated 20.05.2011 or later revisions of the above listed documents approved by EASA. | This approval only addresses the clearance of the pod configurations with respect to flutter, all other aspects of the certification process, such as structural strength<->-<->- effects on aircraft handling and performance, system safety and environmental protection require further investigation. | 30.05.2011 |
| 10043010 | FR AVIATION Ltd. | EASA, A.S.01049 | REV. 1 | | 228-202K | Introduction of ACAS and Mode S | | - Design Change Definition Document -DC-0020 Issue B or later approved revision-AFM supplement-FRAH/FM/0105 Rev 0 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 27.09.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------|----------------|----------------|----------------------|--|---|---------------------|--|---|---------------|
| 10043015 | FR AVIATION Ltd. | EASA.A.S.01558 | | | FALCON 20 BASIC,D AND E SERIES | DC-0038; ACAS II and Mode S Installation | | see FRA Certification Document CP-0038-AFM Supplement FRAH/FM/108 to AFM UK Supplement DTM918 | 1. This STC is approved only for the configuration as defined in DC-00382. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/eng (...) | 02.01.2006 |
| 10043018 | FR AVIATION Ltd. | EASA.A.S.01291 | | | FALCON 20 | FR Aviation Modification Reference DC0032 Introduction of Flight Manual Supplement FRAH/FM/106 | | Approved Flight Manual Reference DTM918 FR Aviation Flight Manual Supplement FRAH/FM/106 Revision 0 dated June 2005 or later approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 05.07.2005 |
| 10043096 | FR AVIATION Ltd. | EASA.A.S.00510 | REV. 2 | | FALCON 20 SERIES D,E AND BASIC | Installation of ProLine IV Integrated Avionics Suite | | Master Document List MDL 1001FRAH/FM/0100 to DTM918, or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 23.02.2006 |
| 10043104 | FR AVIATION Ltd. | EASA.A.S.01714 | | | FALCON 20 BASIC, D AND E SERIE | DC-0052 dtd. 7.6.06: Introduction of TAWS Display | | see Design Change Statement DC-0052 dtd. 7.6.05;-AFM Supplement FRAJFM109 dtd. 19.5.06- associated list of affected technical documents | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall (...) | 09.06.2006 |
| 10043105 | FR AVIATION Ltd. | EASA.A.S.02477 | | | DORNIER 288-202K | Operational Upgrade | | FR Aviation Design Change DC-000 1 Issue 1 or later approved revision.FR Aviation AFM Supplement FRAH/FM104 Revision 0 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall (...) | 24.01.2006 |
| 10043110 | FR AVIATION Ltd. | EASA.A.S.02590 | | | B200 | T AS620 Traffic Advisory System | | FR Aviation Modification DC-0066FR Aviation AFM Supplement FRAH/FM100 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall (...) | 28.07.2006 |
| 10043113 | FR AVIATION Ltd. | EASA.A.S.02710 | | | APPLICABLE TO FALCON 20 BASIC, FAN JET & MYSTERE FALCON 20012 MODFD863 | DC-075 Issue 2 dtd. 8.Mar 07 - Enhanced ATC Mode S or later approved revision: Enhanced ATC Mode S. | | See Design Change Statement DC-0075 dtd. 8. Mar 07 (includes Master Datalist)- Flight Manual Supplement FRAHFM100 Rev. 7 dtd 27.Feb 07 or later approved revision- Instructions for Continued Airworthiness ICA-0075 Rev.A or later approved revision | 1. Certification Basis CS25 initial issue.2. Compliance with TGL 13 and NP A 20-12a has been shown.3. Applicable to Falcon 20 Basic, D & E models Post Mod FD863.4. Not all required Download Parameters/DAP are provided, the AFMLimitation Section contains (...) | 13.03.2007 |
| 10043116 | FR AVIATION Ltd. | EASA.A.S.02794 | | | FAN JET FALCON BASIC, D AND E | DC-0094 Issue 1: Extended RVSM operation<(>-<)> | | 1. DC-0094 Issue 1 dated 11.12.07 or later approved revision;2. FRA AFM Supplement FRAHFM100 revision 9 or later approved revision. | 1. Limited to Fan Jet Falcon with FRA STC EASA.A.S.00510 embodied;2. This STC does not constitute an operational approval;3. Prior to installation of this modification the installer must determine that the interrelationship between this modification a (...) | 08.02.2008 |
| 10043117 | FR AVIATION Ltd. | EASA.A.S.02827 | | | | DC-0088 Rev.1 dtd. 22.12.06: Increase to Max Zero Fuel Weight. | | DC-0088 Iss.1 dtd. 22.12.06 or later approved revision-FRA AFM Supplement FRAHFM100 revision 6 or later approved revision | 1. Limited to Falcon 20 acc. DC-0088 Master Document List Section.2. Cert Basis CAR 4b, and (Elect to Comply to) CS25 Amrdt 2 paragraphs 25.1, 25.25, 25.571<(>-<)>25.1501(b), 25.1519, 25.1581 and 25.1583.3. This STC is approved only for the product co (...) | 05.01.2007 |
| 10043119 | FR AVIATION Ltd. | EASA.A.S.02884 | | | DORNIER 228-202K | Installation of Rockwell Collins TCAS 4000 system including Dual Mode SXPDR. | | - Installation must be in accordance with Master Document List included in the Design Change Statement DC-0090, Iss.1, dated 28th March 2007, or later approved Revisions. - Operation must be in accordance with Pilots Operating Handbook Supplement No.FRAH (...) | 1. Compatibility of this modification with other previously approved modifications must be determined by the installer.2.This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Descrip (...) | 30.03.2007 |
| 10043120 | FR AVIATION Ltd. | EASA.A.S.02956 | | | FAN JET FALCON, FAN JET FALCON | Non Capable Enhanced Mode S CEHS) Transponder Installation.Limited enhanced surveillance capacity restricted to a subset of all required enhanced mode S parameters | | # Design Change Statement DC-0097, Issue 2 or later approved revisions.# Airplane Flight Manual Supplement FRAHFM108Revision 1 or later approved revisions.# Instructions for Continued Airworthiness ICA-0097 issue B or later approved revisions. | 1. Fan Jet Falcon and Fan Jet Falcon Series D aircraft with two mode Stransponders Rockwell Collins TDR-94D on board as per aircraft applicability described in Design Change Statement DC-0097 Issue 2 or later approved revisions.2. JAA Leaflet TGL 13 Cert (...) | 03.07.2007 |
| 10043121 | FR AVIATION Ltd. | EASA.A.S.02957 | | | DORNIER 228-202, SIN 8009,8112 | Enhanced ATC Mode S: Dornier Qualification of existing Elementary Mode S A TC systems to Enhanced Mode S requirements with limited DAP set. | | - Design Change Statement: DC-0096 Issue 3- Ground Test Report: GTR-228-34-0020 Issue B/C- EASA approved Pilot's Operating Handbook Supplement No. FRAHFM/0104 Rev. 1, for Fairchild Dornier 228 with FR Aviation Design Change DC0001, and DC0096, or later a (...) | 1. Restricted to SIN 8009, 8112, and 8115.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other ai (...) | 19.04.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------|---------------------------|----------------|----------------------|---|--|---|--|--|---------------|
| 10043129 | FR AVIATION Ltd. | EASA.A.S.03239 | | | BEECH B200 | RNA V GNSS Non-Precision Approach. | | FR Aviation Design Change DC-O 119.FR Aviation AFM Supplements FRAH/FM/080 Revision 1 dated 23 August 2007andFRAH/FM/0707 Revision 1 dated 23 August 2007. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce noadverse effect upon the airworthiness of the product. The inst (...) | 06.09.2007 |
| 10043132 | FR AVIATION Ltd. | EASA.A.S.03410 | | | BEECH B300 | Introduction of Equipment Associated with Operational Requirements.The introduction of supplementary radio navigation aids to support navigation calibration systemand items of equipment for EU-OPS operation. | | # Installation must be in accordance with Master Document List includedin the Design ChangeStatement DC-0127 Iss.1 dated 28th November 2008 or later ApprovedRevisions.# Operation must be in accordance with Pilots Operating HandbookSupplement No.FRAH (...) | 1. Prior to installation of this modification the installer mustdetermine that the interrelationshipbetween this modification and any other previously installedmodification will introduce noadverse effect upon the airworthiness of the product. The in (...) | 28.11.2008 |
| 10043133 | FR AVIATION Ltd. | EASA.A.S.03411 REVISION 1 | REV. 1 | | BEECH B300 | AD-AFIS Equipment Installation.Installation of equipment tosupport operation as an airfieldcalibration aircraft. | | # Installation must be in accordance with Master Document List includedin the Design ChangeStatement DC-0128 Iss.1: Dated 28th November 2007 or later ApprovedRevisions.# Operation must be in accordance with Pilots Operating HandbookSupplement No.FRAH (...) | 1. Prior to installation of this modification the installer mustdetermine that the interrelationshipbetween this modification and any other previously installedmodification will introduce noadverse effect upon the airworthiness of the product. The in (...) | 09.12.2008 |
| 10043135 | FR AVIATION Ltd. | EASA.A.S.03565 | | | FAN JET FALCON BASIC FAN JET FALCON SERIES D AND E | P-RNAV Compliance Demonstration, as per FR Aviation Design ChangeStatement DC-0133 | | # FR Aviation Design Change Statement DC-0133 Issue 1 - FR Aviation Flight Manual Supplement FRAH/FM/100, Revision 10*- Instructions for Continued Airworthiness ICA-0133, Revision A**Or later EASA Approved revisions of these documents. | 1. Design Change applicable to those Fan Jet Falcon Basic and Fan JetFalcon series D and Eon which FR Aviation STC EASAAS.00510 (Rockwell Collins Proline 4) havebeenpreviously embodied.2. Prior to installation of this modification the installer must (...) | 02.06.2009 |
| 10045438 | FR AVIATION Ltd. | | REV. 1 | | DO228-202K | Introduction of Oil Spill Response Role System | FAR Part 23, Amdt: 1-23The Certification Basis for the original product remains applicable tothis certificate/ approval.The certificated noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | AFMS FRAH/FM/0121 dated 18. June 2013Instruction for Continued Airworthiness ICA-0221 dated 19. June 2013or later revisions of the above listed documents approved by EASA | The approval of this design change is conditional on the large bellyradome being installed. UK CAA Major Modification AAN 23889 refers.S/N: 8115Prior to installation of this modification it must be determined thatthe interrelationship between this m (...) | 08.07.2013 |
| 10045492 | FR AVIATION Ltd. | | | | DORNIER 228-202K | Introduction of Flight Management SystemThis Change replaces the existing R-NAV2 and GPS system with a newFlight Management System, providing an RNPS navigation solution and anoperational B-RNAV approval. Additionally a Multi-Function Display isintr (...) | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | AFMS FRAH/FM/0122 Rev. 0 dated 14.Jun.2013In accordance with Document List in DC-0223, Iss. 1, or laterapproved revision.Instruction for Continued Airworthiness ICA-0223, Iss A, dated17.Jun.2013or later revisions of the above listed documents approved by EASA. | The installation of this modification by third persons is subject towritten permission of the approval holder and disposal of the approvedappropriate documentation. | 28.06.2013 |
| 10042074 | FRAKES AVIATION | EASA.IM.A.S.00228 | | | 1900 1900C 1900D 200 200C 300 300LW B200 B200C B300 B300C | Installation of engine exhaust stub in accordance with Frakes AviationInstallation drawing number FA5726, dated September 6, 1985, for the 200series, and drawing FA5601 dated June19, 1995 for the 300 and 1900series, or later EASA-approvedrevision | | | | 28.10.2004 |
| 10014725 | FRANCE AVIATION | EASA.A.S.02902 | | | BEECH F33 | Installation and BRNAV approval of Garmin GNS-480. | | Installation In accordance with France Aviation Modification ref FA 05/2006, and Technical Specification ref SF-05-2006-001 To be operated in accordance with France Aviation Flight Manual Supplement ref AFM-05-2006-001, Issue 1, dated 3/11/2006 or later approved revisions | | 05.12.2006 |
| 10014839 | FRANCE AVIATION | EASA.A.S.03056 | | | BEECH B200 | Installation of Honeywell MST-67A Mode S Transponder. | | Installation must be in accordance with France Aviation Master Data List Rev. 2, dated 13.06.07 or later Approved Revisions.Operation must be in accordance with France Aviation Aircraft Flight Manual Supplement AFM-01-2007-001 Rev.0 dated 16.04.07 a (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 18.06.2007 |
| 10014914 | FRANCE AVIATION | EASA.A.S.03159 | | | PIPER PA-32 | Rudder Hand Control installation for lower Limbs.The "SD1" system consists in a control for rudder that is handled by the pilot while seated in the LH (Left Hand) station. This "SD1" RHC system is maintained on existing manufacturing rudder system. The system can be easily removed (four screws). | | Master Data List FA 25/2007 rev. 1 Dated 17/07/2007.Flight Manual Supplement AFM-25-2007-001 rev. 1 dated 17/07/2007.Instructions for Continued Airworthiness ICA-25-2007-001 rev. 0 dated 15/06/2007. | | 23.11.2007 |
| 10015002 | FRANCE AVIATION | EASA.A.S.03302 | | | BEECH F90 | Installation of 2 Collins 8.33 KHz COMM System and GTX330D mode S diversity transponder. | | Modification file FA 35/2007. | AFMS No. AFM-35-2007-001, dated 26/09/2007 or later approved revision must be used with the modification. | 10.12.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|----------------------|--|--|---|---|--|---------------|
| 10015014 | FRANCE AVIATION | EASA.A.S.03316 | | | BEECH F90 | IFR/RNAV classification of the GNS530AW coupled to the autopilot; basic IFR classification of the GNS430AW and Mode S transponder GTX330 installation. | | Aircraft Flight Manual Supplement for GTX330/330D, AFM-38-2007-001.Aircraft Flight Manual Supplement for GMX200, AFM-38-2007-002.Aircraft Flight Manual Supplement for GNS530W/AV IFR/ BRNAV/ SIDs/ STARs et approches de non précision AFM-38-2007 (...) | The following AFMS must be used when modification is applied:Aircraft Flight Manual Supplement for GTX330/330D, AFM-38-2007-001 revision 1, dated 10/04/2008, or later approved revisions.Aircraft Flight Manual Supplement for GMX200, AFM-38-2007-0 (...) | 23.05.2008 |
| 10015028 | FRANCE AVIATION | EASA.A.S.03335 | | | PIPER PA-28 | Rudder hand control and brake system for lower limbs disabled pilot. The "LPS" rudder hand control and brake system installation is to permit LLD pilot to operate the aircraft without safety pilot or flight instructor in the right hand pilot station. (...) | | Master data list FA 37/2007, Rev. 2, dated 03/06/2008.Flight Manual Supplement AFM-37-2007-002, Rev. 2, dated 03/06/2008.Instructions for continued airworthiness ICA-37-2007-000, Rev. 0, dated 25/10/2007. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 24.06.2008 |
| 10017205 | FRANCE AVIATION | EASA.R.S.01357 | | | EUROCOPTER DEUTSCHLAND EC-135 | Installation of Avidyne Advisory System TAS-620. | | Definition and Installation:-France Aviation Master Data List ref. FA 03/2007, revision 1, dated 23/03/2007, or later approved revision.-Avidyne TAS-620 Installation manual p/n 32-2351Operation:Flight Manual Supplement ref. AFM-003-2007-001 revisio (...) | Installation limited to EC 135 T2 s/n 274. | 30.11.2007 |
| 10033540 | FRANCE AVIATION | P-EASA.A.S.03527 | | | BEECH A36 | FA 06/2008 Rev. 2 - Avionic change with B-RNAV/SID/STAR/GNSS andconventional non-precision approaches privileges with AP coupling. | | None | None | 06.07.2009 |
| 10015369 | FUEL & UTILITY SYSTEMS GOODRICH COR | EASA.IM.A.S.01007 | | FAA STC ST00174BO | B737-200, B737-200C B737-300, B737-400, B737-500 | EASA Validation of FAA STC ST00174BO for "Installation of GoodrichCorporation Float Switch Transient Suppression Device" on Boeing 737Classic Aircraft | | Master Drawing list for the Float Switch Transient Suppression Device,applicable to Boeing 737-200 through -900 Series Airplanes, Revision F,dated 17th August 2004 or later approved revisionEASA Approval No 2004-11309 for AMOC to FAA AD 2004-15-04 for B737Classic Types | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 17.05.2005 |
| 10016792 | FUGRO AIRBORNE SURVEYS | EASA.IM.R.S.01140 | | TCCA STC SH 98-37 | EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 | Installation of Fugro Airborne Surveys Electro-MagnetometerBoom & Geosurvey Equipment installation | | STC SH 98-37 installation instructions.Installation must bein accordance with DECA Aviation Modification Summary No. MS9805650 Issue 3, dated July 17, 2000 TCCA approved, or later TCCA approved revisions. | 1. STC SH 98-37 Supplemental Flight Manual 4537-01 Revision01 approved 18 July 2000 or later approved revisions.2. DECA Aviation Flight Manual Supplement No. 453707 Revision 1, dated 18 July 2000 or later approved revision3. This STC is approved on (...) | 11.10.2005 |
| 10015328 | G&D AERO PRODUCTS INC | EASA.IM.A.S.00313 | | FAA STC SA3759NM | 35-33, 35-A33, 35-B33, 35-C33, 35-C33A, E33, E33A, F33, F33A, B36TC F33C, H35, K35, N35, P35, S35, V35, V35A, V35B,36, A36, A36TC | Installation of acrylic window liners in accordance with FAA approved Report No1, revision B, dated March 15, 1995 or later EASA-approved revision. | | | 1. This installation should not be incorporated in anyaircraft unless it is determined that the interrelationshipbetween this installation and any previously approved configuration will not introduce any adverse effect upon the airworthiness of (...) | 16.11.2004 |
| 10016428 | G&D AERO PRODUCTS INC. | EASA.IM.A.S.02767 | | FAA STC SA7063NM | BEECH 200 BEECH 300 | Installation of Acrylic window liners | | In accordance with FAA approved drawing lists, drawing lists are detailed in the attached AML to this STC at latest approved revision. | None | 05.03.2008 |
| 10014342 | GAMA ENGINEERING Ltd. | EASA.A.S.01602 | | | BEECH 200 | Major Avionic Upgrade | | Lees Avionics Ltd Modification HLJMOD/673/Lees Avionics Ltd AFM Supplements AFMS673001, AFMS673002, AFMS673003 & AFMS673004 all at Issue 1 datedJanuary 23 2006 or later approved revision. | | 06.04.2006 |
| 10014355 | GAMA ENGINEERING Ltd. | EASA.A.S.01620 | | | BEECH 200, SUPER KING AIR | Major Avionic System Upgrade | | AFMS677001 iss x for GNS530 BRNav complianceAFMS677002 iss x for GNS430 BRNav complianceAFMS677003 iss x for GNX330D Mode S | | 19.12.2005 |
| 10014357 | GAMA ENGINEERING Ltd. | EASA.A.S.01622 | REV. 1 | | BEECH B200C, SUPER KING AIR | Ground Use Battery | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/or emissions levels of the original product areunchanged and remain applicable to this certificate/approval. | GAMA Support Services Programme Reference GSSL-05-012 | None | 10.03.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------|----------------|----------------|----------------------|-------------------------------|---|--|--|--|---------------|
| 10014389 | GAMA ENGINEERING Ltd. | EASA.A.S.01675 | REV. 1 | | BAE.125 SERIES 800 | Title: Upgrade of the dual Flight Management System (FMS). Description of change: The original dual Honeywell FMZ-900 FMS is upgraded to a dual Honeywell FMZ-2000 FMS. | CS-25The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installed in accordance with Gama Engineering Ltd modification HL/MOD/689, reference the Master Documents List (MDL), document reference: MDL-689-01, Issue 1. Other technical documentation: Operation in accordance with Aircraft Flight Manual Supplement (...) | The installer must determine whether this design change is compatible with previously approved modifications. | 05.10.2012 |
| 10014502 | GAMA ENGINEERING Ltd. | EASA.A.S.02570 | | | BEECH 200 | Mode S Transponder Installation. | | Form 1 Modification Statement HL/MOD/706 Issue 4. AFM Supplement AFMS706001 Issue 1. | Beech 200 Series. As per AFMS706001 Issue 1 | 06.02.2008 |
| 10014542 | GAMA ENGINEERING Ltd. | EASA.A.S.02629 | REV. 1 | | A340-311, A340-312, A340-313. | Child Restraint System - CRS - Installation in A340-300 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI F-01, "Child Restraint System". The requirements for environmental protection and (...) | - Certification Plan, ref. doc. No. TN1785-004, Rev. H, dated 15th December 2006- Certification Compliance Check List, ref. doc. No. TN1785-005, Rev. E, dated 8th June 2007- Declaration of Design and Performance, ref. doc. No. DDP-1785-026, Rev. H, dat (...) | Limitations in accordance with § 4 - Limitations of the above mentioned DDP, ref. No. DDP-1785-026. All cabin interiors must be evaluated for compatibility of the Child Restraint System (CRS). The CRS is not approved for aeroplanes that are not defined by § 4 of the above referenced DDP. | 10.12.2012 |
| 10014543 | GAMA ENGINEERING Ltd. | EASA.A.S.02630 | REV. 1 | | A340-643 | Child Restraint System - CRS - Installation in A340-600 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI F-01, "Child Restraint System". The requirements for environmental protection and (...) | - Certification Plan, ref. doc. No. TN1785-004, Rev. H, dated 15th December 2006- Certification Compliance Check List, ref. doc. No. TN1785-005, Rev. E, dated 8th June 2007- Declaration of Design and Performance, ref. doc. No. DDP-1785-026, Rev. H, dat (...) | Limitations in accordance with § 4 - Limitations of the above mentioned DDP, ref. No. DDP-1785-026. All cabin interiors must be evaluated for compatibility of the Child Restraint System (CRS). The CRS is not approved for aeroplanes that are not defined by § 4 of the above referenced DDP. | 10.12.2012 |
| 10014544 | GAMA ENGINEERING Ltd. | EASA.A.S.02631 | REV. 1 | | B747-400 | Installation of MAGE Child Restraint System as detailed in Modification Leaflet for Modification no. B744-1795. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | As defined in Mann Aviation Group (Engineering) Ltd. (MAGE) Modification Leaflet for Modification No. B744-1795 or later revisions of the above listed documents approved by EASA | 1. This STC only covers the installation of the MAGE Child Restraint System on Virgin Atlantic Airways Boeing 747-400 Aircraft, as defined in MAGE DDP1785-026 and in accordance with the instructions and limitations defined in MAGE Crew Manual Supplement (...) | 20.06.2012 |
| 10014626 | GAMA ENGINEERING Ltd. | EASA.A.S.02764 | | | HAWKER 800 XP | Installation of a L-3 Avionics F1000 Flight Data Recorder. | | Installation in accordance with Lees Avionics Ltd Modification HL/MOD/728 Issue 2 or later approved revisions. Operation to be in accordance with Lees Avionics Flight Manual Supplement AFMS728001 Issue 2 for F1000 Flight Data Recorder Installation, or (...) | This Installation is intended for compliance with JAR-OPS 1.715. This STC however grants no operational approval to this installation, that will need to be obtained from the State of registry for the particular aircraft involved as defined in JAR OPS 1.715(g). | 17.10.2006 |
| 10014627 | GAMA ENGINEERING Ltd. | EASA.A.S.02765 | | | BEECH 200 | Enhanced Mode S Upgrade. | | Master Document List MDL730001 Issue 1 AFM Supplements AFMS730001 and AFMS730002. | Beech 200/300 Series AFM Supplements AFMS730001 and AFMS730002. | 12.03.2008 |
| 10014768 | GAMA ENGINEERING Ltd. | EASA.A.S.02960 | | | CESSNA 425 | Installation of Traffic Advisory System | | Lees Avionics Modification HL/MOD/799STC-Definition Document DD 799001 Issue 2 AFM AFMS 799001 Issue 1 | As defined in the AFM supplement | 08.03.2007 |
| 10014769 | GAMA ENGINEERING Ltd. | EASA.A.S.02961 | | | CESSNA 425 | Installation of Garmin GNS 530 & GTX330 | | Lees Avionics Modification HL/MOD/800 STC Definition Document for HL/MOD/800 AFM Supplement AFMS 80001 Issue 1 for GNS 530. AFM Supplement AFMS 80002 Issue 1 for GTX 330 Mode S Transponder. | As per AFM Supplements. | 08.03.2007 |
| 10015050 | GAMA ENGINEERING Ltd. | EASA.A.S.03369 | | | CESSNA 500, CITATION | Upgrade of Mode S Transponder for Enhanced Surveillance | | Lees Avionics Major Change HL/MOD/925 Lees Avionics AFM Supplements AFMS925001 Issue 1 AFMS925002 Issue 1 | | 05.03.2008 |
| 10015153 | GAMA ENGINEERING Ltd. | EASA.A.S.03516 | | | HAWKER HS 125-700 SERIES | Dual FMS Upgrade. | | 1. Master Document List MDL961001 issue 1 2. Modification Statement HL/MOD/961 3. AFM Change Sht AFMS961001 Issue 1 4. AFM Supplement AFMS961002 Issue 1 5. Instructions for Continued Airworthiness ICA961001 issue 1 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. AFM Supplement AFMS961002 Issue 1 | 14.08.2008 |
| 10015186 | GAMA ENGINEERING Ltd. | EASA.A.S.03570 | | | CESSNA 525 | Upgrade of Mode S transponders for enhanced surveillance. | | Lees Avionics Ltd Modification HL/MOD/941 Instructions for Continued Airworthiness ICA941001. Flight Manual Supplement AFMS941001 Issue 1 AFMS941002 Issue 1 AFMS941003 Issue 1 | | 13.06.2008 |
| 10015247 | GAMA ENGINEERING Ltd. | EASA.A.S.03684 | | | HAWKER 700 | Introduction of a Sandel ST3400 Class A terrain awareness warning system. | | Installation in accordance with Change statement: HL/MOD/1000 Form 1. Operation in accordance with Flight manual supplement AFMS 1000-01 Instructions for continued airworthiness reference : ICA1000-01 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 15.12.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------|------------------|----------------|----------------------|--------------------------------------|--|---|---|--|---------------|
| 10017226 | GAMA ENGINEERING Ltd. | EASA.R.S.01395 | REV. 1 | | BO 105 S (CBS-4) | The Pyrotechnic Display System: Carriage and deployment of static flares and roman candle type pyrotechnics, installed each side on external dispenser mounts attached to landing gear skids and interfaced with cabin control box to sequence ignition. Note: (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 1. Flight Manual Supplement GSSL-FMS-BO105-190, issue 1 dated 1 March 2008 or later revisions of the above listed documents approved by EASA.2. Modification GSSL-MAJ-BO105-190, issue 1 dated 12 March 2008.3. Continued Airworthiness Instructions GSSL-CAW-BO105-190, issue 1 dated 20 February 2008 | none | 24.02.2011 |
| 10017247 | GAMA ENGINEERING Ltd. | EASA.R.S.01445 | | | BELL 412 | Installation of a dual electronic horizontal situation indicator (EHSI) system. | | RFMS879001 Issue 1 dated 19 May 2008 or later approved revision # Rotorcraft Flight Manual Supplement HL/MOD/900 Issue 1 or later approved revision # Modification Statement ICA879001 Issue 1 or later approved revision # Instructions for Continued Airworthiness. | 1. The EHSI system must have a minimum software level of Eight (8) installed in the symbol generators.2. The YELLOW symbol generator #5C# flag or display unit #DUE# flag must not be visible prior to departure.3. Autopilot coupled ADF tracking (...) | 22.05.2008 |
| 10027113 | GAMA ENGINEERING Ltd. | P-EASA.A.S.03842 | | | PA-28 | Aspen EFD1000 System | - JAA Interim Policies INT/POL/23/1, INT/POL/23/2 and INT/POL/23/3- FAA AC 23.1309-1D; FAA AC 23.1311-1B; FAA AC 23-23-AMC 20-115B The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate | - Master Data List MDL1059-01 Issue 1- Instructions for Continued Airworthiness ICA1059-01 Issue 1- Flight Manual Supplement AFMS1059-01 Issue 1- Or later revisions of the above listed documents approved by EASA | None | 07.09.2009 |
| 10027314 | GAMA ENGINEERING Ltd. | | | | PIPER PA32 ALL SERIES | Installation of Aspen EFD1000 PFD | CS-23 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. JAA Interim Policies INT/POL/23/1, INT/POL/23/2 and INT/POL/23/3 FAA AC 23.1309-1 (...) | Master Data List MDL1099-01 Instructions for Continued Airworthiness ICA1099-01 Flight Manual Supplement AFMS1099-01 Issue 1 or later revisions of the above listed documents approved by EASA | | 21.09.2009 |
| 10027637 | GAMA ENGINEERING Ltd. | EASA.A.S.03763 | REV. 2 | | F28 MARK 0070 | Interior Refurbishment and System Upgrade | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emission levels of the original product are unchanged and remain applicable to this certificate/ approval. | Change Leaflet 1838 Release A, 15/08/2009 (per original certificate)- Continued Airworthiness TN1838-091 Release B, 15/10/2009 (per original certificate)- Certification Plan TN1838-026 Release A15 December 2009- Flight Manual Supplement FMS1838-095 (...) | None. | 26.11.2012 |
| 10027657 | GAMA ENGINEERING Ltd. | | REV. 1 | | PA-23-235, PA-23-250 PA-E23-25 | Installation of Garmin G500/G600 EFIS PFD/ND | CS-23 The Certification Basis for the original product remains applicable to this certificate/ approval. Non-significant. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are (...) | Master Data List MDL1087-02 Issue 1 Change Statement CS1087-02 Issue 1 Instructions for Continued Airworthiness ICA1087-02 Issue 1 Flight Manual Supplement AFMS1087-02 Issue 1 or later revisions of the above listed documents approved by EASA | None | 21.10.2010 |
| 10027818 | GAMA ENGINEERING Ltd. | | REV. 2 | | PA32 ALL SERIES | Installation of Garmin G500/G600 PFD/ND | CS-23 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. JAA Interim Policies INT/POL/23/1, INT/POL/23/2 and INT/POL/23/3 FAA AC 23.1309-1DF (...) | - Master Data List MDL1100-02 Issue 4- Change Statement CS1100-02 Issue 2- Instructions for Continued Airworthiness ICA1100-02 Issue 1- Flight Manual Supplement AFMS1100-02 Issue 4 or later revisions of the above listed documents approved by EASA | This change is limited to the configuration defined in Change Statement CS1100-02 Issue 2 or later approved revision. | 01.04.2010 |
| 10028029 | GAMA ENGINEERING Ltd. | | | | PA-34-200, PA-34-200T, PA-34-220T | Installation of Aspen EFD1000 PFD as add on | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged (...) | EASA approved Master Document List, Doc. No. MDL1106-01, Issue 1, dated 04-Nov-2009, or later approved issue EASA approved Flight Manual Supplement, Doc. No. AFMS1106-01, Issue 2 dated 04-Nov-2009, or later approved issue Approved Instructions for Cont (...) | Only applicable for airplanes/preconditions as defined Section 1.1, "Installation planning" of Doc. No. CS1106-01, "Change Statement", Issue 2, or later approved issue. For approved Software Level of Aspen EFD1000 refer to "Limitations, Conditions, Exemptions" of Doc. No. CS1106-01, "Change Statement" | 26.11.2009 |
| 10028069 | GAMA ENGINEERING Ltd. | | | | 182 SERIES | Installation of Aspen EFD1000 PFD | CS-23 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. JAA Interim Policies INT/POL/23/1, INT/POL/23/2 and INT/POL/23/3 FAA AC 23.1309-1DF (...) | Master Data List MDL1119-01 Change Statement CS1119-01 Instructions for Continued Airworthiness ICA1119-01 Flight Manual Supplement AFMS1119-01 Issue 1 or later revisions of the above listed documents approved by EASA | None | 30.11.2009 |
| 10029002 | GAMA ENGINEERING Ltd. | | | | PA-31 | Installation of defence aid antennas and systems | CS-23 The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | - Master Data List MDL1116-01 Issue 2- Change Statement CS1116-01 Issue 2- Instructions for Continued Airworthiness ICA1116-01 or later revisions of the above listed documents approved by EASA | None | 25.02.2010 |
| 10029092 | GAMA ENGINEERING Ltd. | | | | PA-31 | ATC transponder upgrade for Mode S enhanced surveillance | CS-23 The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | - Master Data List MDL1120-01 Issue 1- Change Statement CS1120-01 Issue 1- Flight Manual Supplement AFMS-1120-01 Issue 1- Instructions for Continued Airworthiness ICA1120-01 Issue 1 or later revisions of the above listed documents approved by EASA | Not Applicable | 03.03.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------|------------------|----------------|----------------------|--|---|--|--|---|---------------|
| 10029406 | GAMA ENGINEERING Ltd. | | REV. 1 | | 1. 35-33, 35-A33, 35-B33 2. 35-C33, 35-C33A, E33, E33A 3. E33C, F33, F33A, F33C, H35 4. J35, K35, M35, N35, P35 5. S35, V35, V35A, V35B 6. 36, A36, A36TC, B36TC 7. 95-55, 95-A55, 95-B55 8. 95-C55, D55, E55, 58, 58P 9. 58TC | Amendment to existing STC 10029406 to add TAWS-B functionality and include optional GAD43 autopilot interface unit. | CS-23The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval | Master Data List MDL1112-02 Issue 1Change Statement CS1112-02 Issue 1Instructions for Continued Airworthiness ICA1112-02 Issue 1Flight Manual Supplement AFMS1112-02 Issue 1or later revisions of the above listed documents approved by EASA | None | 03.01.2011 |
| 10029499 | GAMA ENGINEERING Ltd. | | | | BEECH 90 | Navigation/ communications systems upgrade. | CS-23The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.AMC20-4AMC20-5AMC20-13The certificated noise and/ or emissions levels of the orig (...) | MDL1130-01 Issue 2CP-1130-01 Issue 1IP-1130-01 Issue 1ICA-1130-01 Issue 1AFMS-1130-01 Issue 1AFMS-1130-02 Issue 2AFMS-1130-03 Issue 2or later revisions of the above listed documents approved by EASA. | None. | 29.03.2010 |
| 10029724 | GAMA ENGINEERING Ltd. | P-EASA.A.S.03861 | REV. 1 | | PA-28 | Installation of Garmin G500/G600 EFIS PFD/ND. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS-23.EASA GM GA/G001 has been applied.The certificated noise and/ or emissions level (...) | Master Data List MDL1050-02 Issue 1Change Statement CS1050-02 Issue 1Instructions for Continued Airworthiness ICA1050-02 Issue 1Flight Manual Supplement AFMS1050-02 Issue 1or later revisions of the above listed documents approved by EASA | This change is limited to the configuration defined in Change StatementCS1050-02 Issue 1 or later approved revision. | 15.04.2010 |
| 10029726 | GAMA ENGINEERING Ltd. | | REV. 1 | | PIPER PA-28 | | | | | 16.04.2010 |
| 10030465 | GAMA ENGINEERING Ltd. | P-EASA.R.S.01596 | | | BO105 | Mounting boom for camera installation.Certification of the installation of a mounting boom which can be usedfor installation of a camera. The camera installation provisions and themaximum weight and frontal area of cameras which can be used with the (...) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalThe applicant has shown compliance with the current EASA certificationspecification CS-2 (...) | Rotocraft Flight Manual Supplement: RFMS 1089-01, Issue 1.Instructions for Continued Airworthiness: ICA 1089-01, Issue1Installation/ Post-Installation Procedures: IPI 1089-01, Issue 1Master Document List: MDL 1089-01, Issue 1or later revisions of the above listed documents approved by EASA | CAT-A-operation is prohibited (CAT-A-operation is excluded fromcertification).Maximum weight for cameras used is 43.1 kg and maximum frontal area is 1470 sqcm.The used camera has to be certified separately. | 16.06.2010 |
| 10030863 | GAMA ENGINEERING Ltd. | P-EASA.A.S.03823 | | | EXTRA EA-300 / L | External camera installation | The Certification Basis for the original product remains applicable tothis certificate/ approval | Master Documents List Drg No MDL 1054-01 Issue 1ICA-1054-01 Issue 1 Instructions for Continued AirworthinessAFMS-1054-01 Flight Manual Supplementor later revisions of the above listed documents approved by EASA | None | 09.07.2010 |
| 10031729 | GAMA ENGINEERING Ltd. | | | | PIPER PA-34 ALL SERIES | Installation of Garmin G500/G600 PFD/ND | The Certification Basis for the original product remains applicable tothis certificate/ approvalCS 23Non-significantTherequirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the originalproduct are (...) | Master Data List MDL1101-01 Issue 1Change Statement CS1101-01 Issue 1Instructions for Continued Airworthiness ICA1101-01 Issue 1Flight Manual Supplement AFMS1101-01 Issue 1or later revisions of the above listed documents approved by EASA | This change is limited to the configuration defined in Change StatementCS1101-01 Issue 1 or later approved revision. | 08.09.2010 |
| 10031769 | GAMA ENGINEERING Ltd. | | REV. 2 | | SEE EASA APPROVED MODEL LIST | Child Restraint System - CRS - Installation in Various Aircraft. Rev. 2; Child Restraint System - CRS - Installation in Additional Aircraft. | In accordance with the EASA approved TCDS ListThe Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicabletothis certificate/ approval- CRI D-0321-001, "Child Restraint System (...) | - Certification Proposal Document CPD-CRS-01, Rev. 1, dated 15th January 2013; - Certification Plan, ref. doc. No. TN 1898-005, Rev. B, dated 7thSeptember 2010; - Declaration of Design and Performance - DDP, ref. doc. No.DDP-1898-012, Rev. A, dated 10th (...) | Limitations in accordance with § 4 - Limitations of the above mentioned DDP, ref. No. DDP-1898-012.All cabin interiors must be evaluated for compatibility of the ChildRestraint System (CRS). The CRS is not approved for aeroplanes that arenot defined (...) | 24.01.2013 |
| 10032795 | GAMA ENGINEERING Ltd. | | REV. 1 | | 112, 112TC, 112B, 112TCA, 114, 114A, 114B, 114TC | Aspen EFD1000 Software Version 2.3.1 upgrade | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS-23, Amendment2The requirements for environmental protection and the associatedcerti (...) | Flight Manual Supplement AFMS-1108, Issue 2Instructions for Continued Airworthiness ICA-1108, Issue 2or later revisions of the above listed documents approved by EASA. | None | 03.05.2012 |
| 10033225 | GAMA ENGINEERING Ltd. | | | | PA-44-180 & PA-44-180T | Installation of Garmin G500/G600 PFD/ND | CS-23The Certification Basis for the original product remains applicable tothis certificate/ approvalThe requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval | Master Data List MDL1151-01 Issue 1Change Statement CS1151-01 Issue 1Instructions for Continued Airworthiness ICA1151-01 Issue 1Flight Manual Supplement AFMS1151-01 Issue 1or later revisions of the above listed documents approved by EASA | None | 05.01.2011 |
| 10033770 | GAMA ENGINEERING Ltd. | P-EASA.A.S.03723 | | | CESSNA 310 | Introduction of EXP5000 electronic primary flying display and dualGNS430W GPS navigation/communication units. | | Lees Avionics Limited Change HL/MOD/1031Flight Manual Supplements: AFMS1031-01 Issue 1 and AFMS1031-02 Issue 1 | None | 27.07.2009 |
| 10033778 | GAMA ENGINEERING Ltd. | P-EASA.A.S.03724 | | | CESSNA 310 | Introduction of a GDC31 autopilot roll steering converter | | Lees Avionics Limited Change HL/MOD/1032Flight Manual Supplement AFMS1032-01 Issue 1 | None | 27.07.2009 |
| 10034337 | GAMA ENGINEERING Ltd. | | | | SEE EASA APPROVED MODEL LIST | Navigation/Communications Upgrade. | The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Master Data List MDL-1208 Issue 1Change Statement CS1208 Issue 1Instructions for Continued Airworthiness ICA-1208 Issue 1Flight Manual Supplement AFMS-1208-01 Issue 1Flight Manual Supplement AFMS-1208-02 Issue 1Flight Manual Supplement AFMS-1208-03 Is (...) | None | 25.03.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------|---------------------|----------------|----------------------|---|--|---|--|---|---------------|
| 10034531 | GAMA ENGINEERING Ltd. | | | | SEE EASA APPROVED MODEL LIST | Installation of Garmin G500 EFIS PFD/ND. | CS-23 Amendment 1The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Master Data List MDL-1211 Issue 1Change Statement CS1211-01Issue 1Instructions for Continued Airworthiness ICA-1211-01Issue 1Flight Manual Supplement AFMS-1211-01 Issue 1 for later revisions of the above listed documents approved by EASA. | None. | 11.04.2011 |
| 10034814 | GAMA ENGINEERING Ltd. | | | | SEE EASA APPROVED MODEL LIST | Navigation/ communications upgrade | CS-23The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to thiscertificate/ approval | Master Data List MDL-1224 Issue 1Change Statement CS-1224 Issue 1Instructions for Continued Airworthiness ICA-1224 Issue 1Flight Manual Supplement AFMS-1224-01 Issue 1Flight Manual Supplement AFMS-1224-02 Issue 1Flight Manual Supplement AFMS-1224-03 (...) | None | 03.05.2011 |
| 10037701 | GAMA ENGINEERING Ltd. | | | | SEE EASA APPROVED MODEL LIST | GNS-W Navigation System Qualification | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalCS-23 Amendment 2The requirements for environmental protection and the associatedcertif (...) | Gama Engineering Limited Change Statement CS-1259 Issue 2 dated 12Dec11Master Data List MDL-1259 Issue 2 dated 12Dec11Instructions for Continued Airworthiness ICA-1259 Issue 2 dated 12Dec11Flight Manual SupplementsAFMS-1259-01 Issue 2 dated 12Dec11A (...) | None | 16.12.2011 |
| 10037800 | GAMA ENGINEERING Ltd. | | | | A) BN2A-8, BN2A-9, BN2A-20 A) BN2A-21, BN2A-26 A) BN2A-27, BN2B-20, BN2B-21 A) BN2B-26, BN2B-27, BN2T A) BN2T-4R, BN2T-4S B) BN2A MARK III-3 B) BN2A MARK III, III-1, III-2 | GNS-W Navigation System Qualification in accordance with MDL-1275, Issue2. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/approval:CS23 initial issue:23.1, 23.1301, 23.1309(a), (b), (d), (e), 23.1357, 23.1367(a),23.138 (...) | AFMS-1275-01, Issue 2AFMS-1275-02, Issue 2or later revisions of the above listed documents approved by EASAINstallationInstruction IPI-1275, Issue 2Instructions for Continued Airworthiness ICA-1275, Issue 2 | None | 22.12.2011 |
| 10039502 | GAMA ENGINEERING Ltd. | | | | BN2A MARK III TRISLANDER, BN2A MARK III-1 TRISLANDER, BN2A MARK III-2 TRISLANDER, BN2A MARK III-3 TRISLANDER. | Installation of Aspen EFD1000 Electronic Primary Flight Display | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS23 Amendment 2The requirements for environmental protection and the associatedcertif (...) | Master Data List MDL-1282 Issue 1, dated 04 May 2012; Change Statement CS-1282 Issue 1, dated 04 May 2012; Flight Manual Supplement AFMS-1282 Issue 1 or later revisions of the above listed documents approved by EASA | None | 08.05.2012 |
| 10041163 | GAMA ENGINEERING Ltd. | | | | P2006T | Wingtip pyrotechnics & LED lights. | CS - 23 Issue 2The Certification Basis for the original product remains applicable tothis certificate/ approval.CS - 23as per TCDS EASA.A.185 Major. | Certification Plan - CP-1305 issue 1;Substantiation report SUB-1306 issue 1;Flight test Report FTR-1306;Master Documents List - MDL-1306;Flight Manual Supplement AFMS-1306;Instructions for Continued Airworthiness ICA-1306;or later revisions of the above listed documents approved by EASA. | As per: Flight manual supplement AFMS - 1306 & Instructions for ContinuedAirworthiness ICA-1306. | 24.08.2012 |
| 10043560 | GAMA ENGINEERING Ltd. | | | | 206A, 206B, 206L-1, 206L-3, 206L-4. | Sweep 20 Infrared Sensor Suite Installation | The Certification Basis for the original product as amendedby thefollowing additional or alternative airworthiness requirements: CS 27Amendment 2 (as in CP-1281 Issue B)The requirements for environmental protection and the associatedcertificated noise (...) | Rotorcraft Flight Manual Supplement RFMS-1281-01 Issue 2or later revisions of the above listed documents approved by EASA- Master Drawing List MDL-1281 Issue 2, dated February 04,2013- Instructions for Continued Airworthiness ICA-1281, Issue 1 | Prior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noadverse effect upon the airworthiness ofthe produc (...) | 07.02.2013 |
| 10044260 | GAMA ENGINEERING Ltd. | | REV. 1 | | B200, B200C, B200GT B200CGT, B300, B300C | | The Certification Basis for the original product as amendedby thefollowing additional or alternative airworthiness requirements:Special Condition(s): SC-F23.1353-02 (as per CRI F-58)CS 23 amndt.3 paragraphs mentioned in doc CP-1312The requirements fo (...) | According to GAMA Master Documents List MDL-1312 is.1 dated15-Mar-2013;To be operated in accordance with GAMA AFM-S Document FCS-1312 is.1dated 14-Mar-2013To be maintained i.a.w.GAMA ICA Document ICA-MDB05-01 is.1, dated15-Mar-2013 or later revisio (...) | According limitation section of AFM-SPrior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviouslyinstalled design change and/ or repair will introduce noadverse effect (...) | 08.04.2013 |
| 10015372 | GARMIN INTERNATIONAL, Inc. | EASA.IM.A.S.01023 | | SA01254W1 | DIAMOND DA-40 | Installation of Garmin G1000 in the DA40 and DA40F airplane | | Master Drawing List, 005-C0004 Rev U or later approved revision-Required Equipment List, 005-00305 Rev K or later approved revision-EASA Airplane Flight Manual Supplement, 190-00303-81 Rev 1 or later approved revision-G1000 Maintenance Manual, 190-00545-00 or later approved revision | 1. Hartzell Propeller STC EASA.IM.A.S.01644 must be installed.2. Diamond Aircraft Industries Modifications listed in Garmin Drawing 005-00304-00 Rev 10 or later approved revision must be installed | 11.08.2006 |
| 10033158 | GARMIN INTERNATIONAL, Inc. | P-EASA.IM.A.S.03191 | REV. 3 | FAA STC SA01535W1-D | BEECH 200, 200C, B200, B200C BEECH 300, 300LV, B300, B300C | Installation of the Garmin G1000 Integrated Avionics Systemand GFC 700Automatic Flight Control System and optional Electronic Stability andProtection system (ESP). | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.(See EASA approved ANNEX)The requirements for environmental protection and the associa (...) | Model 200 SeriesInstallation in accordance with Garmin Master Drawing List (MDL)005-00421-00, Revision 3, dated 3/14/2009Operation in accordance with Garmin Airplane Flight ManualSupplement(AFMS) 190-00915-04, Revision 1, dated 3/14/2009 and EASA Ap (...) | Additional airframe, engine and propeller configuration requirements aredefined in the data shown on the Garmin Master Drawing List (MDL)identified above.Aircraft modified by thisSTC must be maintained in accordance with theinstructions for Continue (...) | 04.07.2013 |
| 10034918 | GARMIN INTERNATIONAL, Inc. | | | FAA STC SA01595W1-D | G36 | G1000 Integrated Avionics System update with options including Synthetic/Vision and Pathways and GTS 820 Traffic Advisory SystemEASA Approval of SBAS based LNAV/VNAV and LPV Functionality. | The Certification Basis for the original product remains applicable tothis certificate/ approval | Installation in accordance with Garmin Master Drawing List (MDL)005-00620-18, Revision 1, dated July 21, 2010.Operation in accordance with Garmin FAA Approved Flight ManualSupplement 190-01258-00, Revision 1, dated July 21, 2010, and EASAspecific Fli (...) | None | 13.05.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|--|---|--|---|--|---------------|
| 10034986 | GARMIN INTERNATIONAL, Inc. | | | FAA STC SA01584W1-D | G58 | G1000 Integrated Avionics System update with options including Synthetic Vision and Pathways and GTS 820 Traffic Advisory System. EASA Approval of SBAS based LNAV/VNAV and LPV Functionality. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Garmin Master Drawing List (MDL) 005-00606-01, Revision 1, dated June 4, 2010. Operation in accordance with Garmin FAA Approved Flight Manual Supplement 190-01180-01, Revision 1, dated June 4, 2010, and EASA specific Flg (...) | None | 18.05.2011 |
| 10037574 | GARMIN INTERNATIONAL, Inc. | | REV. 1 | FAA STC SA02019SE-D | SEE EASA APPROVED MODEL LIST | Installation of Garmin GTN 6XX/7XX Navigation System | As defined by FAA STC SA02019SE-D. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Installation in accordance with Configuration 3 and 4, as applicable of Garmin Master Drawing List (MDL) 005-00533-C0, at Revision 8, dated 31 July 2013 and Garmin Installation Checklist ref 190-01007-E1, at Revision indicated in the MDL. Earlier Revi (...) | In order to approve adaptations for each specific installation a Minor Change must be raised for each installation as defined in the Garmin Installation Checklist 190-01007-E1, at Revision indicated in the MDL identified above. Prior to installation of (...) | 07.08.2013 |
| 10015734 | GARRETT AVIATION SERVICES | EASA.IM.A.S.01757 | | FAA STC ST301CH | CESSNA 500, 550, S550, 560 | Validation of FAA STC ST301CH - Installation of a Honeywell CAS-67A Traffic Alert and Collision Avoidance System (TCAS II) in accordance with ElectroSonics Top Drawing ES550-100116-01, Rev G, dated January 14, 2004 or later approved revision | | AFM supplement ES550-100116-101 Revision C (FAA approved) for STC ST301CH (AFMS E550-200156-18A) ElectroSonics drawings: ES500-200116-156 and -316. Instructions for continued airworthiness: ICA ES550-200116-200A & ES550-200116-201A Technical Investiga (...) | Note: This STC complies with requirements for Mode S elementary surveillance. Limitations & Conditions as on FAA STC | 13.11.2006 |
| 10014086 | GATE V AIRCRAFT MAINTENANCE GmbH | EASA.A.S.01177 | | | CESSNA 550 | Integration of a TCAS and Upgrade of Mode S Transponders to Elementary Surveillance | | - 5597/04/1-12 Instructions for Continued Airworthiness - 5597/04/1-06 Airplane Flight Manual Supplement | | 10.03.2005 |
| 10015409 | GENERAL AEROSPACE INC | EASA.IM.A.S.01123 | | FAA STC ST02902AT | BOEING 767-200 SERIES BOEING 767-300 SERIES | Installation of a honeywell RESCU 406 AF ELT system | | General Aerospace Master Drawing List GA672-ELT-MDL revision D dated April 19, 2005 or later FAA approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 30.05.2005 |
| 10015450 | GENERAL AEROSPACE INC | EASA.IM.A.S.01194 | REV. 1 | FAA STC ST03042AT | BOEING 767-200 BOEING 767-300 | Installation of Honeywell Aircraft Communication Addressing & Reporting System (ACARS). | | General Aerospace Master Drawing List No. GA67-ACARS-MDL Rev. B | | 12.06.2008 |
| 10015451 | GENERAL AEROSPACE INC | EASA.IM.A.S.01195 | | FAA STC ST03041AT | BOEING 767-200 SERIES BOEING 767-300 SERIES | Installation of a Honeywell ARINC Data Acquisition Management System | | General Aerospace Master Drawing List No. GA67-DFDR-MDL revision B dated May 7, 2005 or later approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 30.05.2005 |
| 10015470 | GENERAL AEROSPACE INC | EASA.IM.A.S.01223 | | FAA STC ST03002AT | BOEING 767-200 BOEING 767-300 | Installation of a Honeywell Enhanced Ground Proximity Warning System. | | General Aerospace Master Drawing List No. GA672-EGPWS-MDL Revision B dated May 24, 2005 or later EASA approved Revisions. General Aerospace AFM Supplement GA67-EGPWS-AFMS01 dated December 10, 2004 or later EASA approved revisions. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 14.06.2005 |
| 10015475 | GENERAL AEROSPACE INC | EASA.IM.A.S.01228 | | FAA STC ST00115AT | BOEING 767-200 BOEING 767-300 | Installation of Dual Mode S Transponders & Honeywell Traffic Alert & Collision Avoidance System II - FAA STC ST00115AT | | General Aerospace Inc Engineering Reports GA-34-25 Rev B dated 5 August 1993 and GA-GE52/0704/ER Rev B dated 24 May 2005 and Master Drawing List GA-34-1023 Rev C dated 24 May 2005, or later EASA approved revisions. General Aerospace Inc. AFM Supplements (...) | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 14.05.2005 |
| 10015620 | GENERAL AEROSPACE INC | EASA.IM.A.S.01572 | | FAA STC ST03138AT | BOEING 737-200 BOEING 737-300 BOEING 737-400 BOEING 737-500 | Installation of Honeywell Enhanced Ground Proximity Warning System according to FAA STC ST03138AT | | General Aerospace Master Drawing List No. GA37-EGPWS-MDL, dated 20 June 2005 Airplane Flight Manual Supplement, Ref. GA37-EGPWS-AFMS01, dated 22 July 2005 FAA STC ST03138AT, issued 22 July 2005 | According to FAA STC ST03138AT | 03.03.2006 |
| 10015639 | GENERAL AEROSPACE INC | EASA.IM.A.S.01602 | | FAA STC ST03126AT | BOEING 737-100 BOEING 737-200 BOEING 737-300 BOEING 737-400 BOEING 737-500 | Installation of a honeywell RESCU 406AF Emergency locator transmitter (ELT) System according to FAA STC ST03126AT | | General Aerospace Master Drawing List Ref. FA373-ELT-MDL, dated 7 May 2005 | According to FAA STC ST03126AT | 29.11.2005 |
| 10016072 | GENERAL AEROSPACE INC | EASA.IM.A.S.02253 | | FAA STC ST01792AT | BOEING 757-200 | TCAS and Dual Mode-S Transponders Installation. | | General Aerospace Master Drawing List GA-RC02/898/MDL, 12th September 1998 AFM Supplement GA-RC02/898/AFMS 31st December 1998. | | 17.10.2007 |
| 10016466 | GENERAL AEROSPACE INC | EASA.IM.A.S.02812 | | FAA STC ST03486AT | BOEING 737-200 BOEING 737-300 BOEING 737-400 BOEING 737-500 | Installation of a Honeywell ARINC 717 Flight Data Acquisition Management System (FDAMS). | | FAA letter of compliance with EASA Certification Basis dated 12 March 2008. Master Drawing List No. GA37-DFDAU-MDL01, Revision A, dated 24 April 2007 or later revision of the aboves listed documents approved by EASA in accordance with EASA ED decision (...) | | 19.03.2008 |
| 10035023 | GENERAL AEROSPACE INC | EASA.IM.A.S.02759 | | FAA STC SA3169SO | B737-200, B737-300 B737-400, B737-500 | Installation of a Collins Traffic Alert and Collision Avoidance System and Dual Mode-S Transponders. | As defined in TCDS | General Aerospace Master Drawing List GA-34-1002 Rev. EFAA approved AFM Supplement GA135AFM dated NOV 23, 1993 or FAA approved AFM Supplement GA136AFM dated SEP 15, 1998 and EASA AFM Supplement GA37-ELS/EHS-AFMS01 Issue iRor later revision of the abov (...) | A copy of this STC must be included in the permanent record of their craft. | 30.07.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------|-------------------|----------------|----------------------|---|---|---------------------|---|--|---------------|
| 10016948 | GENEVA AVIATION, INC. | EASA.IM.R.S.01343 | REV. 1 | FAA STC SH4747NM | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 D | Installation of consoles, toggle switches, circuit breakers, and replacement wiring harnesses in accordance with MDL GA107. This STC is the validation of FAA STC SH4747NM (issued on July 26, 1989 and last amended August 8, 2008). | | 1. Master Drawing List GA107 Rev. M dated September 25, 2008 or later EASA approved revision.2. Rotorcraft Flight Manual Supplement GA107-2, Revision R dated 9/25/08 or later EASA approved revision.3. Rotorcraft Flight Manual Supplement GA107- (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 25.11.2008 |
| 10016960 | GENEVA AVIATION, INC. | EASA.IM.R.S.01357 | | FAA STC SR00474SE | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 C EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 | Modular Equipment Rack installation per MDL GA179. | | Master Drawing List GA179 Rev. B dated November 20, 2002 or latter approved revisions Rotorcraft Flight Manual Supplement GA179-3 Revision D dated 2/15/07 or latter approved revisions Compliance Checklist CCL-179 Rev -- dated July 5, 2007 or latter approved revisions | This STC provides approval for structural provisions only. A separate approval is required for the installation of any equipment in the modular equipment rack. | 03.04.2008 |
| 10016961 | GENEVA AVIATION, INC. | EASA.IM.R.S.01359 | | FAA STC SR00326SE | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 C EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 | Microwave Pod installation per MDL GA170 | | Master Drawing List GA170 Rev. C dated May 30, 2008 or later approved revisions Rotorcraft Flight Manual Supplement GA170-3 Rev. B dated May 14, 2008 or later approved revisions Compliance Checklist CCL-170 Rev. -- dated July 18, 2007 or later approved revisions | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 15.07.2008 |
| 10016973 | GENEVA AVIATION, INC. | EASA.IM.R.S.01374 | | FAA STC SR00222SE | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 C EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 | Installation of a supplemental avionics console per MDL GA158. | | Master Drawing List GA 158 dated 03/02/95 Rev. Level or later approved revisions. Compliance Checklist CCL-158 Rev. -- Dated August 16, 2007 or later approved revisions. | This STC provides approval for structural provisions only. A separate approval is required for the installation of any equipment in the supplemental avionics console. | 07.03.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------|-------------------|----------------|----------------------|--|---|---------------------|---|---|---------------|
| 10016974 | GENEVA AVIATION, INC. | EASA.IM.R.S.01375 | | FAA STC SR00242SE | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 C EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 | Installation of a new instrument panel and pedestal per MDLGA162. | | Master Drawing List GA162 at Rev. Level B dated 11/15/96 or later EASA approved revisions, Installation Instructions GA162-2 Rev -- dated 8/23/95 or later EASA approved revisions, Compliance Checklist CCL-162 Rev -- dated August 14, 2007 or later EASA approved revisions | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 11.12.2008 |
| 10016975 | GENEVA AVIATION, INC. | EASA.IM.R.S.01376 | | FAA STC SH5119 NM | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 C EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 EUROCOPTER FRANCE AS 355N | Rear Bench Seat Modification per MDL GA141 | | Master Drawing List GA141 Rev. C dated 3/22/95 or later approved revisions Rotorcraft Flight Manual Supplement GA141-3 Revision C dated January 19, 2001 or later approved revisions Compliance Checklist CCL-141 Rev - dated August 17, 2007 or later approved revisions | | 03.04.2008 |
| 10016992 | GENEVA AVIATION, INC. | EASA.IM.R.S.01401 | | FAA STC SR00388SE | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 C EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 | Installation of a rear communications console per MDL GA178 This STC is the validation of FAA STC SR00388SE (last amended January 18, 2000) | | Master Drawing List GA178 dated 10/31/96 Rev. Level A or later approved revisions Rotorcraft Flight Manual Supplement GA178-3 dated 4/25/03 Rev. Level B or later approved revisions Compliance Checklist CCL-178 Rev A dated January 16, 2008 or later approved revisions | This STC provides approval for structural provisions only. A separate approval is required for the installation of any equipment in the communication console. The rear communications console installation is not approved for the cabin configuration with the three seat rear bench. | 06.03.2008 |
| 10017002 | GENEVA AVIATION, INC. | EASA.IM.R.S.01413 | | FAA STC SR00361SE | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 C EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 EUROCOPTER FRANCE AS 355N | Utility Rack installation per MDL GA164. | | Master Drawing List GA164 Rev. B dated May 16, 2008 or later approved revisions Rotorcraft Flight Manual Supplement GA164-3 Rev. # initial release dated May 15, 2008 or later approved revisions Compliance Checklist CCL-164 Rev. B dated February 15, 2008 or later approved revisions | This STC provides approval for Utility Rack installation only. A separate approval is required for the installation of any equipment in the utility rack. Prior to installation of this modification the installer must determine that the interrelati (...) | 16.07.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------|---------------------|----------------|----------------------|--|--|---|--|--|---------------|
| 10017004 | GENEVA AVIATION, INC. | EASA.IM.R.S.01415 | | FAA STC SH5120NM | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 C EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 EUROCOPTER FRANCE AS 355N | Installation of a two place front seat. | | GA142-3 revision C dated 7 July 2000 or later approved revision # Flight manual Supplement; GA142-2 revision G dated 16 November 1995 or later approved revision # Installation Instructions; GA142 revision K dated 29 July 1995 or later approved revision # Top Drawing List; | Copilot dual controls may not be installed with this STC; Number of occupants increased to 7 (1 pilot and 6 passengers) with this STC installed. The following placard to be installed on the instrument panel in front of the pilot: AIRCRAFT IS BALLASTED (...) | 20.03.2008 |
| 10017054 | GENEVA AVIATION, INC. | EASA.IM.R.S.01483 | | FAA STC SR00191SE | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 C EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 EUROCOPTER FRANCE AS 355N | Installation of a single place pilot/copilot seat in accordance with MDL GA156 | | Master Drawing List GA156 Rev. B dated 23 August, 2007, or later approved revisions Compliance Checklist CCL-156 Rev - dated May 6, 2008 or latter approved revisions | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 27.08.2008 |
| 10030234 | GENEVA AVIATION, INC. | P-EASA.IM.R.S.01482 | | FAA STC SH4776NM | AS 350 B, AS 350 B1, AS 350 B2 AS 350 B3, AS 350 BA, AS 350 D | Installation of replacement transmission cowling panels i.a.w. Master Drawing List GA119. This STC is the validation of FAA STC SH4776NM (issued on October 06, 1989). | | - Master Drawing List GA119 Rev. B dated 8/17/07 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | 1., Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2., T (...) | 17.02.2009 |
| 10030479 | GENEVA AVIATION, INC. | P-EASA.IM.R.S.01510 | | FAA STC SR00219SE | 1. AS350B, AS350B1, AS350B2 1. AS350BA, AS350D 2. AS355E, AS355F, AS355F1, 2. AS355F2, AS355N | Horizontal Stabilizer Color Camera Mount This STC is the validation of FAA STC SR00219SE (last amended August 11, 1995) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Drawing List GA161, Rev. Level: -; dated 21/07/1995, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 17.06.2010 |
| 10036806 | GENEVA AVIATION, INC. | P-EASA.IM.R.S.01362 | REV. 1 | FAA STC SR00521SE | SEE EASA APPROVED MODEL LIST | Installation of Geneva Aviation Audio System. The purpose of this revision is to update the applicability of this approval to new helicopter types and to validate FAA STC SR00521SE last amended on January 23rd, 2013. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Type AS350; Master Drawing List GA182 Rev L dated December 3rd, 2012; Flight Manual Supplement GA 182-3 Rev / dated June 6th, 1998; Instruction for Continued Airworthiness ICA182-2 Rev. A dated October 15th, 2012; Rotorcraft Type AS355; Master Draw (...) | The applicability of this STC does not include the model AS350B3 incorporating modification OP-4305 (Arriel 2D engine installation). | 12.09.2013 |
| 10036868 | GENEVA AVIATION, INC. | P-EASA.IM.R.S.01400 | | FAA STC SR00643SE | 1. AS350 D/-B/-B1/-B2/-BA 2. AS355 E/F/-F1/F2 3. BO-105 A/-C/-S/-LS A-1 3. BO-105 LS A-3 4. BELL 206 L-3/L-4 4. BELL 206A/-A-1/-B/-L/L-1 4. BELL 407 | Installation of Geneva Aviation Video System per Geneva Aviation Master Drawing List GA183 This STC is the validation of FAA STC SR00643SE (last amended April 17, 2009) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Modification complies with FAR 27 at amdt. level as specified in Compliance Checklist C (...) | Definition and installation: Geneva Aviation "VIDEO SYSTEM" Master Drawing List GA183 Rev K Geneva Aviation "VIDEO SYSTEM Installation Instructions" GA183-2 Rev F (for AS350/AS355 model aircraft) Geneva Aviation "VIDEO SYSTEM Installation Instructions to (...) | Any optional Geneva Aviation modifications listed in Installation Instructions, to be installed in conjunction with this STC (->->-) shall be EASA approved. This STC covers installation of the complete set of equipment as defined by this STC or a subset of (...) | 13.10.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|-------------------|----------------|--------------------------------|--|--|--|--|--|---------------|
| 10016721 | GEOTECH LTD | EASA.IM.R.S.01016 | | | AS 350 SERIE | Survey System Installation (Structural & Electrical Power provisions) | | -DECA Aviation Engineering Ltd Modification Summary MS04193Revision 1 dated December 2, 2004 or later approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 31.05.2005 |
| 10015647 | GKN AEROSPACE CHEM-TRONICS Inc. | EASA.IM.A.S.01620 | | | AIRBUS A318 AIRBUS A319 AIRBUS A320 AIRBUS A321 | Replacement CFM56-5B Inlet Barrel Lower Panel Assembly (Part Number 030-13001-1)Replacement CFM56-5B Inlet Barrel LowerPanel Assembly (Part Number 030-13001-1) | | Master Drawing List ref MDL030-13014 revision D, dated April 20, 2005 or later approved revision | | 30.01.2006 |
| 10016008 | GKN AEROSPACE CHEM-TRONICS Inc. | EASA.IM.A.S.02173 | | FAA STC ST01189LA | BOEING DC-9-81/82 BOEING DC-9-83 (MD-83) BOEING DC-9-87 (MD-87) BOEING MD-88 | Installation of ASTECH Manufacturing, Inc. inlet assembly | | - ASTECH Manufacturing , Inc. Document No. MDL-030-06009 titled: "Master Drawing List - MD-80 Series Inlet", Revision A00, datedJune 20, 2001, or later FAA approved revision | | 31.10.2006 |
| 10016312 | GKN AEROSPACE CHEM-TRONICS Inc. | EASA.IM.A.S.02616 | | FAA STC SA5818NM | MD-88 | FAA STC SA5818NM #Installation of Astech/MCI tailpipe assembly. | | Astech/MCI doc. no. ER-92-716 #Drawing list - MD 80 Series tailpipe, Rev N/C dated July 10, 1992 or later FAA approved revision. | This approval should not be extended to other specified airplanes of this model series on which other previously approved modifications are incorporated, unless it is determined that the interrelationship between this change and any those other prev (...) | 19.03.2008 |
| 10016441 | GKN AEROSPACE CHEM-TRONICS Inc. | EASA.IM.A.S.02786 | | FAA STC ST01898LA; WILLIAM BON | AIRBUS A319 AIRBUS A320 AIRBUS A321 | Installation of ASTECH Engineered Products fan cowl door in accordance with ASTECH#s Master Drawing List V2500 Cowl Door Assembly No. MDL030-16103, Revision #A8. | | # FAA STC ST01898LA# Master Drawing List V2500 Cowl Door Assembly Drawing List ref. No. MDL030-16103, Revision A, dated April 18th 2006 or later revisions approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequentrevisions o (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.1 (...) | 10.10.2008 |
| 10016571 | GKN AEROSPACE CHEM-TRONICS Inc. | EASA.IM.A.S.02969 | | FAA STC ST01578LA;BOND TEL.011 | AIRBUS A319-113 AIRBUS A319-114 AIRBUS A320-111 AIRBUS A320-211 AIRBUS A320-212 | Installation of ASTECH Engineered Products, Inc. Inlet Inner BarrelAssembly in accordance with Master Drawing List, A320/CFM56-5ASeries Engine Inlet, Drawing List Number MDL030-11030 Revision C. | | # FAA STC ST01578LA# Master Drawing List A320 CFM56-5A Series Engine Inlet, ref. No. MDL030-11030, Revision C, dated 2 April 2008 or later revisions approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of th (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 12.12.2008 |
| 10014305 | GKN AEROSPACE SERVICES LIMITED | EASA.A.S.01549 | REV. 1 | | BOEING 737 | Number 1 Cockpit Windshields. Addition of layers of polyurethane interlayer to the existing qualified designs to improve in-service reliability | | GKN Aerospace Modification C/R 014. | This approval provides for the installation of the referenced windows individually or in any combination. | 12.01.2007 |
| 10014677 | GKN AEROSPACE SERVICES LIMITED | EASA.A.S.02838 | | | BOEING 747 | Introduction of cockpit windows with improved design for Boeing 747 series aircraft. GKN p/n 06314 and 06315 as replacement for Boeing p/n 65B27042 and 65B27046 and GKN p/n 06316 and 06317 as replacement for Boeing p/n 065B27043 and 65B27047. | | 1. B747 No. 2 and 3 supplement IPC.2. B747 No. 2 and 3 window ACMM. | This approval provides for the installation of the referenced windows individually or in any combination. | 10.04.2007 |
| 10014782 | GKN AEROSPACE SERVICES LIMITED | EASA.A.S.02980 | | | BOEING 737-100 BOEING 737-200 BOEING 737-200C BOEING 737-300 BOEING 737-400 BOEING 737-500 BOEING 737-600 BOEING 737-700 BOEING 737-700C BOEING 737-800 BOEING 737-900 | Number 2 Cockpit Windshields Addition of Layers of Polyurethane Interlayer to the Existing Qualified Design to improve In-Service Reliability. | | This approval provides for the installation of the referenced windows individually or in any combination. | 19.12.2007 | |
| 10036545 | GKN AEROSPACE SERVICES LIMITED | P-EASA.A.S.03492 | REV. 1 | | B777-200, B777-200LR B777-300, B777-300ER B777F | Installation of a GKN Aerospace designed Boeing 777 No 1 Windshield PartNo. 10702 (LH) / 10703 (RH); or Part No. 10712 (LH) / 10713 (RH) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.-CS 25 Amdt 4-CR1 H-1 - IA on EWIS not applicable per date of application toEASA | -Major Change Approval C/R 017 Iss. 1 dtd. 19 July 2011.-ICaper GKN Component Maintenance Manual 56-11-07or later revisions of the above listed documents approved by EASA | None | 02.11.2011 |
| 10044546 | GLOBAL AEROSPACE DESIGN | | | FAA STC ST03183CH | B757-200/-300 | Installation of the Thales ISFD System(Description: Installation of the Thales Integrated Standby FlightDisplay System in accordance with Global Aerospace Design CorporationMasterDrawing List GR0002-MDL Revision C, dated January 15, 2013 or later FAA approved revision.) | The Certification Basis (CB) for the original product remains applicableto this certificate/ approval.The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Master Drawing List, Global Aerospace Document No. GR0002-MDL RevisionC-(s,-s)- dated: 15 January 2013-Instructions for Continued Airworthiness, Global Aerospace DocumentNo. GR0010-ICA, Revision A, dated: 10 September 2012- Airplane Flight Manual (...) | Prior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noadverse effect upon the airworthiness ofthe product (...) | 19.04.2013 |
| 10015318 | GLOBAL AIRWORKS, Inc. | EASA.IM.A.S.00254 | REV. 1 | FAA STC ST01626LA; EASA.IM.A.S | BOMBARDIER CL-600-2B19 BOMBARDIER CL-600-2C10 BOMBARDIER CL-600-2D24 | Installation of Airworks Cockpit Door Surveillance SystemRevision 1: Amended FAA STC ST01626LA | | FAA STC No. ST01626LA dated 30 October 2006.FAA SoC reference 103L-08; 13 dated 22 December 2008.AirWorks Index List Document No. IL23002, Revision A, dated 3 March 2004 or later approved revision.AirWorks Airplane Flight Manual Supplement Document N (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 05.01.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|---------------------|----------------|--------------------------------|--|--|---|---|--|---------------|
| 10016684 | GLOBAL AIRWORKS, Inc. | EASA.IM.A.S.C.01059 | REV. 1 | FAA STC ST01585LA | A330-201, A330-202, A330-203 A330-223, A330-243, A330-301 A330-321, A330-322, A330-323 A330-341, A330-342, A330-343 | Installation of AirWorks Cockpit Door Surveillance Systems (Validation of the FAA STC ST01585LA). This STC installs a CDSS. The Revision 1 has been issued to correct the references of Associated Technical Documents. | The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Index Document List (IDL) No. IL23009 revision D, dated 11 October 2009 Aircraft Flight Manual Supplement, ref: AFMS230019 Revision A, dated 24 March 2005 Instruction for Continued Airworthiness, ref: ICA230078, dated 20 August 2003 Instruction for Contin (...) | Prior to the installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product. | 30.11.2011 |
| 10041546 | GLOBAL AIRWORKS, Inc. | | | FAA STC ST3042CH | ATR 42-200, ATR 42-300, ATR 42-320, ATR 42-500 ATR 72-101, ATR 72-102, ATR 72-201, ATR 72-202, ATR 72-211, ATR 72-212, ATR 72-212A. | Installation of a Global AirWorks Cockpit Door Surveillance System. This STC installs a cockpit door surveillance system (CDSS) allowing the crew to view video images of the area outside the cockpit door. | The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Standard Aero Master Data List, ref: MDL230071, Rev F., dated 01 February 2012 Standard Aero Aircraft Flight Manual Supplement, ref: AFMS230071, Rev A, dated 14 February 2012 Instruction for Continued Airworthiness, ref: ICA 230071, Rev B Standard Aer (...) | None | 26.09.2012 |
| 10015704 | GODERICH AIRCRAFT Inc. | EASA.IM.A.S.01691 | | | BEECH B300 | Artex ELT C406-N Installation. | | Avionics Design Services Ltd. Master Drawing List MDL05096 Revision N/C dated 16 September 2005 or later approved revision. Avionics Design Services Ltd. For Continued Airworthiness ICA05096-1 Revision N/C dated 15 September 2005 or later approved revision. | Modification limited to s/n FL-175. Maintenance must be in accordance with above Continued Airworthiness document | 17.01.2006 |
| 10015705 | GODERICH AIRCRAFT Inc. | EASA.IM.A.S.01692 | | | BEECH B300 | Universal UNS 1E Flight Management System Installation. | | Avionics Design Services Ltd. Master Drawing List MDL05097 Revision N/C dated 21 September 2005 or later approved revision. Avionics Design Services Ltd. For Continued Airworthiness ICA05097-1 Revision N/C dated 21 September 2005 or later approved revision. | Modification limited to s/n FL-175. Operation must be in accordance with Goderich Aircraft Inc. Flight Manual Supplement FMS05097-1 Revision N/C approved 20 September 2005 or later approved revision. Maintenance must be in accordance with above Continued Airworthiness document. | 17.01.2006 |
| 10015886 | GODERICH AIRCRAFT Inc. | EASA.IM.A.S.02004 | | O-LSA06-081/D (LSTC) | BEECH 400A (HAWKER 400 XP) | Cabin Entertainment Equipment Installation. | | Cabin Entertainment Equipment Installation according to Master Drawing List MDL06025 Rev. N/C, dated 30 March 2006 or later approved revision. Maintenance must be in accordance with Avionics Design Services Ltd. Instructions for Continued Airworthiness I (...) | This installation is limited to the Serial No. RK-37 (Registration SE-DRS). This approval is for electrical/avionics aspects only. Structural aspects of this installation are not included and require separate approval. | 09.11.2006 |
| 10015887 | GODERICH AIRCRAFT Inc. | EASA.IM.A.S.02005 | | O-LSA06-117/D (LSTC) | BEECH 400A (HAWKER 400 XP) | Change in interior seating arrangement, addition of sink/toiletting rear lavatory, fabrication and installation of RH fwd cabinet, revision of existing LH fwd cabinet, revision to interior finishing details, and addition of new fuselage antenna. | | Installation according to Phoenix Aero Aviation Engineering Ltd. Engineering Service Bulletin PA-ESB-06-002 Issue 1, Rev. Original approved 28 April 2006 or later approved revision. Operation must be in accordance with Phoenix Aero Aviation Engineering (...) | 1. This installation is limited to the Serial No. RK-37 (Registration SE-DRS). 2. Compatibility of this modification with other previously approved modifications must be determined by the installer. | 19.12.2006 |
| 10016452 | GODERICH AIRCRAFT Inc. | EASA.IM.A.S.02798 | | TCCA STC O-LSA08-056/D ISSUE 3 | FALCON 10 | EASA Validation of Transport Canada Limited STC O-LSA08-056/D Issue 3: Cabin Entertainment Equipment Upgrade for Falcon 10 serial number 161 aircraft in accordance with Avionics Design Services Ltd. Master Drawing List MDL08009 Revision B, dated 31st (...) | | See those of original Transport Canada Limited STC O-LSA08-056/D Issue 3 Avionics Design Services Ltd. Master Drawing List MDL08009 Revision B, Issue 1, dated 31st March 2008. Goderich Aircraft Inc. Maintenance Manual Supplement MMS-PA-08-009 Revisio (...) | EASA STC limited to Falcon 10 serial number 161 aircraft. See original Transport Canada Limited STC O-LSA08-056/D Issue 3. See Avionics Design Services Ltd. Master Drawing List MDL08009 Revision B, Issue 1, dated 31st March 2008. Goderich Aircraft Inc. Ma (...) | 21.04.2008 |
| 10014214 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.01374 | REV 1 | | CESSNA 182E CESSNA 182F CESSNA 182G CESSNA 182H CESSNA 182J CESSNA 182K CESSNA 182L CESSNA 182M CESSNA 182N CESSNA 182P CESSNA 182Q CESSNA F182-P CESSNA F182-Q | Installation of Noise Reduction Kit C182-606650 | | - Anhang zum Flughandbuch SPOH_12498_I0550_HM- Anhang zum Flughandbuch SPOH_12498_I0550_NQ- Anhang zum Flughandbuch SPOH_12498_O550_EH- Anhang zum Flughandbuch SPOH_12498_O550_NQ- Umrüstungsanweisung II_12498 | 1. The approval is valid only in combination with the installation of FAA STC SA009133SC (Texas Skyways Inc) or STC SA00152WI (Air Plains Services Corp.) Both with own EASA approval. 2. The approval for the models Cessna 182 E, F and G is not v (...) | 04.08.2006 |
| 10014302 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.01546 | REV. 1 | | DR300/108, DR300/120 DR300/125 DR300/140, DR300/180R DR315, DR340, DR360, DR380 DR400/120, DR400/120A DR400/120D, DR400/125 DR400/140, DR400/140B DR400/160, DR400/160D DR400/180, DR400/180R DR400/180S DR400/2+2, DR400NGL, DR400/100 | DR300/400 MOGAS Operation Installation of a modification kit to allow the operation with MOGAS i.a.w. EN228 to aircrafts models and variants as listed above. The installation can be performed to aircrafts of the existing fleet and to aircrafts production. | The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | General: Master Document List MDCL_2030326_rev1_20130815 Model DR300) and DR315 - no wing tanks, retrofit Flight Manual Supplement SPOH_8696_DR300_rev1_20130415 Installation Instruction II_2030326_DR300_RF_FT_rev0_20130415 Maintenance Manual Supplemen (...) | According limitation section of AFMM must be maintained iaw. Original MM and in addition iaw Supplemental Maintenance Manual MMS_2030326_rev0_20130415 or later approved revisions. Prior to installation of this modification it must be determined that the in (...) | 18.10.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|----------------|----------------|----------------------|--|--|---|--|---|---------------|
| 10014483 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.02546 | | | PIPER PA-23-250 | Modification 13452Alteration of Piper PA-23-250 aircraft to aerial work with a Leica ADS 40 digital camera installation in addition to the installation of Zeiss RMK 15 Top Cameras based on EASA approval No. 2004-9428 | | Certification Program CP 13452 REV0 22/11/2005Compliance Record CR 13452 REV0 02/03/2006Master Drawing List MDRAWL_13452REV0_28/02/2006Supplement to POH SPOH 13452 REV0 22/11/2005Installation Instruction II 13452 REV0 20060224.docStructural Report SR 13452 REV1 20060223.mcd | | 13.03.2006 |
| 10014590 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.02699 | | | DIAMOND HK-36 SUPER DIMONA | Modification of the Dimona HK 36 R "Super Dimona" equipped with the engine Rotax 912 A3 with the alternative Propeller MTV-21-A-CF/C175-05. | | Service Bulletin: GFM SB 2008-2Anhang zum Flughandbuch für den Motorsegler HK 36 R SPOH_14326_rev0_23/08/2007 Ausgabe 08.2007.Wartungshandbuch für den Motorsegler HK 36 R_30204-r13-HK36-WHB, Ausgabe 25. Juni 1990Umrüstungsanweisung, Modification 14326 | All S/N of the variant: HK 36 R "Super Dimona" | 11.06.2008 |
| 10014635 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.02779 | | | CESSNA 172S | Installation of Noise Reduction Kit C172S-606560 | | - Anhang zum Flughandbuch Lärminderungskit C172S-606560-Umrüstungsanweisung 14768 | None | 26.09.2006 |
| 10014670 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.02829 | REV. 3 | | 172RG | Modification 14839Installation of Low Noise Silencer Kit "System Gomolzig" C172RG_606520Additional Propeller MTV | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels. | Low Noise Silencer Kit C172RG_606520Master Document List MDOCL_14839_rev2_20130601Flight Manual Supplement SPOH_14839_rev2_20130601 (German version) or SPOH_14839_rev2_20130601 (English version) Installation Instruction II_14839_rev2_20130601 or I (...) | According to limitation section of AFM. Must be maintained in accordance with original Cessna MM and in addition law Supplemental Maintenance Manual SMM_14839_rev1_20130601 or later approved revisions. Prior to installation of this modification it must be determined that (...) | 03.09.2013 |
| 10014832 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.03047 | REV. 1 | | 1. PA-34-200T (SENECA II) 1. PA-34-220T (SENECA III) Version 2024425 Camera Hatch and Equipment Installation 2. PA-34-220T (SENECA V) | System PA-34-812000Installation of Penta Camera System System PA-34-813000 | The Certification Basis for the original product remains applicable to this certificate/ approval. FAR 23 law, TCDS EASA, I.M.A. 090 and FAA A7SOT. The requirements for environmental protection and the associated certificated noise and/or emissions levels o (...) | System PA-34-812000Flight Manual SPOH_14043_rev0_06112006Installation Instruction II 14043_rev0_06112006System PA-34-813000Master Document List MDOCL_2024425_rev0_20111201Flight Manual Supplement SPOH_2024425_rev0_20110915Installation Inst (...) | As referred in AFM-S | 21.12.2011 |
| 10014836 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.03052 | | | CESSNA 182 | Towing Kit Cessna 182 "System Gomolzig" C182-820000. The Cessna 182 and its series equipped with a towing system for banning towing and sailplanes up to 850 kg MTOW. | | Anhang zum Flughandbuch Modifikation 15530, Towing Kit "System Gomolzig" C182-820000, Ausgabe: 08.07 Doc-No: SPOH_15530_rev1_16/08/2007. Umrüstungsanweisung Modifikation 15530 Towing Kit "System Gomolzig" C182-820000, Ausgabe: 08.07 Doc-No.: II_15530_rev0_23/05/2007. | Cessna 182M, N.P.Q.R., equipped with SMA SR 305-23-MA01 Diesel engine according to STC EASA.A.S.00774. | 12.10.2007 |
| 10014907 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.03145 | REV. 1 | | TP206C-E, TU206B, TU206E-G, T206H. | Cessna Exhaust System with Extension 1. TP206C-E, TU206B, E-G Exhaust System with Extension c)206E(-)6065802, T206H Low Noise Silencer Kit CT206H-606570 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 1. TP206C-E, TU206B, E-G Exhaust System with Extension C)206E(-)606580MDOCL_2028022_rev0_20121001Installation Instructions II_2028022_rev0_20121001AFM-Supplement: SPOH_2028022_rev0_20121001To be maintained in accordance with law MMS_2028022_rev0_201210012.T206H Low (...) | as referred in limitations section of the AFM-S xxx | 30.11.2012 |
| 10015022 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.03328 | | | CESSNA 401 CESSNA 401A CESSNA 401B CESSNA 402 CESSNA 402A CESSNA 402B CESSNA 402C CESSNA 411 CESSNA 411A CESSNA 414 CESSNA 414A CESSNA 421 CESSNA 421A CESSNA 421B CESSNA 421C CESSNA 425 | Alternative Supplemental Inspection Document (A-SID). | - According to Gomolzig MDL Document No. MDOCL17070_REV0_04122008 or later EASA approved Revisions. - To be operated and maintained in accordance with approved Gomolzig Document No. A7CE_A-SID_Master Manual_17070, Original Issue or later EASA approved Revisions. | 1. Alteration to A-SID is subject to written permission of approval holder. 2. This modification is only limited to S/N listed in EASA approved Gomolzig Document No. 17070_Master Aircraft Application Control List_01122008_rev0, or later EASA a (...) | 05.12.2008 | |
| 10015027 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.03334 | | | CESSNA 208B | Installation of Camera Kit Cessna 208B "System Gomolzig" C208B-812000 as defined by Equipment List AEL_17000_Rev0_11112007.XLS, dated 11.11.2007 | | - Flight Manual Supplement SPOH_17000_rev1_12/22/2007.DOC, dated 12.12.2007 or later approved revision. Installation Instruction "Umrüstungsanweisung", II_17000_rev1_13/12/2007.DOC, dated 13.12.2007 or later approved revision | | 19.12.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|------------------|----------------|----------------------|--|---|---|---|--|---------------|
| 10015031 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.03343 | REV. 2 | | DO 28 D-2 | Turbine Conversion of Do 28 D-2 (model marketing name: Do 28-G92 and Do28-G92-EF) including new Propeller variants Master Drawing List G92 for STC-Version G92-A-1900, G92-EF-A-1900, G92-A-1950 and G92-EF-A-1950 Master Document List MDOCL_2023625_rev0_2 (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels. | i.a.w. STCDS EASA 10015031 Section A - H Chapter IV Operating and Service Instructions 1. Flight Manual: DORNIER Do 28G92 Airplane Flight Manual issued for Z/N4134, approved November 1996 (HgCAA), including approved AFM-Supplements (HgCAA), or later app (...) | According limitation section of AFM DO28G92, Maintenance Manual Supplement MMS_DO28-G92_rev0_20130115 and STCDS 10015031 | 10.04.2013 |
| 10015034 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.03347 | REV 1 | | BEECH 99 BEECH B99 | Installation of P&W PT6A-34 Engine | | Flight Manual Supplement SPOH_08959_15022008_rev 0 , dated 15.02.2008 Installation Instructions II_08959_15022008_rev0, dated 15.02.2008 | This STC is limited to Beech 99 / B99 models equipped with Propeller McCauley 4HFR34C769/94LMA-4 i.a.w. FAA STC SA1351GL. | 19.12.2008 |
| 10015040 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.03353 | | | LEARJET 35A LEARJET 36A | Installation of Laser Reflector Dekagon | | As per Master Document List MDOCL_17150, rev. 0 (or later approved revisions) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 05.08.2008 |
| 10015121 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.03470 | | EASA STC 2004-1522 | CESSNA 402 CESSNA 402A CESSNA 402B CESSNA 402C | Use of Aerial Survey Camera Kit C402-812000 for installation of Zeiss RMK Top 15 and Top 30 cameras. Note: Change to the existing EASA STC 2004-1522 to enable optional operation of digital camera systems or laser scanners in accordance with MDL MDRWL_14631_revC_10102007. | | Flight Manual Supplement to the Pilot Operating Handbook - Aerial Survey Camera Kit C402-812000 Rev. D, dated 25/01/2008. Installation and Maintenance Instructions: Umrüstanweisung Kamerarüstsatz System Gomolzig C402-812000 Rev. C, dated 25/01/2008. | | 08.02.2008 |
| 10015261 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.03719 | | LBA STC 0279/2038 | BRITTEN NORMAN BN-2A-21 ISLAND BRITTEN NORMAN BN-2A-26 ISLAND BRITTEN NORMAN BN-2A-27 ISLAND BRITTEN NORMAN BN-2B-20 ISLAND BRITTEN NORMAN BN-2B-26 ISLAND | Installation of Exhaust System Gomolzig BN2-606500. | | - According to Gomolzig Master Document List MDOCL_19057_rev0_27112008, or later Revisions. To be operated and maintained in accordance with approved Gomolzig Documents Flight Manual Supplement SPOH_19057_rev0_27112008 and Installation Instruct (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pro (...) | 04.12.2008 |
| 10017100 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.R.S.01056 | | | HUGHES 369D, HUGHES 369E | Installation of Isolator Washing Platform | | - Flight Manual Supplement SPOH_12155, dated 31.07.2006 or later EASA approved revision: Umrüstanweisung (Installation Manual) II_12155 | Helicopter must be equipped with High Skid Landing Gear and McDonnell Douglas Helicopter Cargo Hook Kit P/N 369H90072-505 | 06.12.2006 |
| 10027347 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | P-EASA.A.S.03887 | REV. 1 | | SA226T | Installation of MX20 Camera system. Installation of Wescam MX series (10, 15, 20) Gimbal Cameras with manual extension/retraction mechanism, operator console and downlink antenna. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Document List MDOCL_19719_rev1_20130501- AFM Supplement SPOH_19719_rev1_20130501- Installation Instruction II_19719_rev1_20130501 or later revisions of the above listed documents approved by EASA. | According limitation section of AFM. Aircraft must be previously modified with camera hatches i.a.w. Swearingen Camera Hole Modification. Prior to installation of this design change it must be determined that the interrelationship between this design cha (...) | 29.05.2013 |
| 10029335 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | | | | 310, 310A, 310B, 310C, 310D, 310E, 310F, 310G, 310H, 310I, 310J, 310J-1, 310K, 310L, 310N 310P, 310Q, 310R E310H, E310J T310P, T310Q, T310R | Alternative Supplemental Inspection Document (A-SID) | CS-23.B The Certification Basis for the original product remains applicable to this certificate/ approval | According to Gomolzig MDL Document No. MDOCL21530_REV0_15122009 To be operated and maintained in accordance with approved Gomolzig Document No. 3A10_A-SID_Maintenance Manual Supplement_21530, Original Issue CP_21530_rev_0_15122009 or later revisions of the above listed documents approved by EASA. | As referred in limitations section of the 3A10_A-SID_Maintenance Manual Supplement_21530 Alteration to A-SID is subject to written permission of approval holder. This modification is only limited to S/N listed in EASA approved Gomolzig Document No. 21530 (...) | 19.03.2010 |
| 10029380 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | | | | CESSNA 182, -A, -B, -C, -D, -E CESSNA 182F, -G, -H, -J, -K, CESSNA 182L, -M, -N, -P, -Q, CESSNA 182R, -S, T, T182 CESSNA T182T | Camera Modification Kit | CS-23.C The Certification Basis for the original product remains applicable to this certificate/ approval | - Master Document List MDOCL_2022177_rev0_01022010-Airplane Flight Manual Supplement SPOH_2022177_rev0_01022010 for Cessna 182S, T and T182T-Anhang zum Flughandbuch Kamerarüstsatz C182-812000 Ausgabe 21.03.97 for Cessna 182, -A, -B, -C, -D, -E, -F, -G, -H, -J, -K, -L, -M, -N, -P, -Q, -R, T182 | As referred in limitations section of the AFM-S-Airplane Flight Manual Supplement SPOH_2022177_rev0_01022010 for Cessna 182S, T and T182T-Anhang zum Flughandbuch Kamerarüstsatz C182-812000 Ausgabe 21.03.97 for Cessna 182, -A, -B, -C, -D, -E, -F, -G, -H, -J, -K, -L, -M, -N, -P, -Q, -R, T182 | 19.03.2010 |
| 10029570 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | | | | HUSKY A-1, A-1A, A-1B, A-1C | Husky A1 Silencer STC Ext. To Variant A, B and C | CS-23.C The Certification Basis for the original product remains applicable to this certificate/ approval | - Master Document List MDOCL_21750_rev0_20100301-Airplane Flight Manual Supplement SPOH_21750_rev0_20100301- Installation Instruction II_21750_rev0_20100301 or later approved revisions of the above listed documents | As referred in limitations section of the AFM-S-Airplane Flight Manual Supplement SPOH_21750_rev0_20100301 | 01.04.2010 |
| 10029736 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | P-EASA.A.S.03144 | | | CESSNA 421B | Installation of Silver Jodide Generators on Cessna 421B | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval FAR 23, amendments 1 through 7 (not required by 21A.101, elected to comply by the applic (...) | - Flight Manual Supplement SPOH_16351_rev1_15032009 (dated 15.03.2009)- Master Document List MDOCL_16351_rev1_12042010 or later revisions of the above listed documents approved by EASA | Not Applicable | 16.04.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|----------------------|----------------|--------------------------|------------------------------|---|--|--|--|---------------|
| 10030084 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | LBA STC 0060/549B | REV. 1 | | SEE EASA APPROVED MODEL LIST | Upwards Opening Door for Parachutist Dropping. This STC revises and supersedes LBA STC 0060/549b. | The Certification Basis for the original product remains applicable to this certificate/ approval. CAR 3. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and (...) | Airplane Flight Manual Supplement SPOH_2022824_rev0_15042010 Installation Instruction II_2022824_rev0_15042010 or later approved revisions of the above listed documents | As referred in limitations section of the AFM-SSPOH_2022824_rev0_15042010 For parachutist dropping | 08.10.2010 |
| 10030085 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | LBA STC 0059/549B | REV. 1 | | SEE EASA APPROVED MODEL LIST | Dropping of max 4 Parachutists with Manual or Automatic Release. This STC revises and supersedes LBA STC 0059/549B. | The Certification Basis for the original product remains applicable to this certificate/ approval. CAR 3. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and (...) | Airplane Flight Manual Supplement SPOH_2022926_rev0_15042010 Installation Instruction II_2022926_rev0_15042010 or later approved revisions of the above listed documents | As referred in limitations section of the AFM-SSPOH_2022926_rev0_15042010 Max 4 parachutists | 08.10.2010 |
| 10031342 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | | | | 208B | C208B Ferry Fuel Tank Installation in accordance with Gomolzig Flugzeug-und Maschinenbau GmbH MDOCL_2023163, Rev. 0, dated 01/08/2010 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Flight Manual Supplement (Modification 2023163), Edition 01/06/2010 Installation Instruction (Modification 2023163), Edition 01/06/2010 Maintenance Instruction included in Installation Instruction or later revisions of the above listed documents approved by EASA | none | 11.08.2010 |
| 10038775 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | | REV. 1 | M6/91 | PA-25-235, PA-25-260 | Low Noise system PA-25-235-606500 Modification 2027423 applicable to PA-25-235 and PA-25-260. Low Noise system PA-25-235-606560 Modification 2027423 Variant B applicable to PA-25-235 only | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The following paragraph(s) at a later amendment: CS-36 Amdt. 2 and FAR 36 Amdt. 28. This certificate/ approval involves a change to the requirements (...) | Low Noise system PA-25-235-606500 Modification 2027423 applicable to PA-25-235 and PA-25-260. According to Gomolzig Master Document List No. MDOCL_2027423_20120201 and Gomolzig Installation Instructions Doc. No. II_2027423_rev0_20120201, or later Rev (...) | According limitation section of AFM-S. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 12.03.2013 |
| 10040526 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | | | | PA-31T | PA-31T Camera Hole Modification Change Note CN_2025593_rev0_20120201 | The Certification Basis for the original product remains applicable to this certificate/ approval. FAR 23 Amdt 1-7. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and (...) | MDOCL_2025593_rev0_20120201 Installation Instructions II_2025593_rev0_20120201 To be operated in accordance with SPOH_2025593_rev0_20120201 To be maintained in accordance with II_2025593_rev0_20120201 or later revisions of the above listed documents approved by EASA | as referred in AFM-S | 09.07.2012 |
| 10040826 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | 0010016836-001 | | LBA STC EMZ SA 0100/2025 | CESSNA 340A | 1. Camera Modification Kit System C340-8120002. C340 Camera-latch Installation System C340-812500 Change Note CN_2027313_rev0_20120402 | The Certification Basis for the original product remains applicable to this certificate/ approval. FAR 23 Amdt 1-7. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and (...) | 1. System C340-812000AFM-Supplement: Luftbildversion Gomolzig C340-812000 Ausgabe 27.12.1988 Installation Instruction: Umrüstanweisung zum Kameraausrüstungssatz C340-812000 Ausgabe 12/882. System C340-812500 MDOCL_2027313_rev0_20120402 Installation Instruct (...) | As referred in AFM | 27.07.2012 |
| 10042992 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.0080 | REV. 1 | | DR 400 RP | Installation of engine Lycoming O-540 A4D5 or O-540 J3A5 | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels. | "Flughandbuch DR 400 RP mit Motor O-540 A4D5/J3A5, Ausgabe 08.04.2002"; "Anhang zum Flughandbuch DR 400 RP, S/N 1766 mit Motor O-540 J3A5", Doc. No. SPOH_08834_rev0_20121201, doc. later revisions of the above listed documents approved by EASA-Modifica (...) | None | 22.03.2013 |
| 10043326 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | EASA.A.S.03863 | | | CESSNA 340 CESSNA 340A | Alternative Supplemental Inspection Document (A-SID) | | - According to Gomolzig MDL Document No. MDOCL20100_REV0_30042009 or later EASA approved Revisions - To be operated and maintained in accordance with approved Gomolzig Document No. 3A2S_A-SID_Master Manual_20100_rev0 or later EASA approved Revisions | 1. Alteration to A-SID is subject to written permission of approval holder. 2. This modification is only limited to SIN listed in EASA approved Gomolzig Document No. 20100_Master Aircraft Application Control3_Ust_30042009_rev0 or later EASA approved R (...) | 01.07.2009 |
| 10043650 | GOMOLZIG FLUGZEUG- UND MASCHINENBAU | | | | O-540-J3A5 | Use of Automotive Gasoline RON 98 per EN228 for Lycoming Engine O-540-J3A5 (S/N L24488-40A) | FAR 33 Amendment 18 for the following affected areas: 33.1 Applicability, 33.3 General, 33.5 Instruction manual for installing and operating the engine, 33.7 Engine ratings and operating limitations, 33.2 Engine cooling, 33.35 Fuel and induction system (...) | Instructions for operation, maintenance and inspection: Gomolzig Supplement to Lycoming Operator's and Maintenance Manual SOM2027338, initial issue; or later revisions of the above listed documents approved by EASA. | The data in the FAA TCDS E-295 remains valid, except the following: Fuels: Aviation Gasoline, 100, 100LL or 100VLL per ASTM D910, Automotive Gasoline RON 98 per EN228 (max 1% alcohol), see Gomolzig Supplement to Lycoming Operator's and Maintenance Manual (...) | 15.02.2013 |
| 10016207 | GOODRICH CORPORATION | EASA.I.M.A.S.02452 | | FAA STC ST02551CH | BOEING 747-400F | Installation of Centerline Restraint System on the main cargo deck of a B747-400F in accordance with Master Data List MDL 7M39300. Validation of FAA STC ST02551CH. | | MD2522 Weight and balance control and loading manual supplement. MD2512 Aircraft cargo loading manual supplement. MD2535 Aircraft maintenance manual supplement. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 02.07.2008 |
| 10026859 | GOODRICH CORPORATION | P-EASA.I.M.A.S.03106 | | FAA STC ST00989SE-D | B767-300F | Installation of the CHS (Cargo Handling System) on the B767-300FMD: MDL7N10000FAA STC: ST00989SE-D | The Certification Basis has been determined and is defined in CRI A-01. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The Certification (...) | FAA: or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | The operation of the STC is only approved when installed on the provisions of the Boeing A/C B767-300F. | 18.08.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|----------------------|---|--|---|--|--|---------------|
| 10031947 | GOODRICH CORPORATION | | REV. 1 | TCCA STC SA10-3 | DHC-8-401, DHC-8-402 | Installation of Goodrich De-Icer boots on the Aircraft Wings, Horizontal Stabilizer and Vertical Stabilizer FAA STC ST02634NY, reissued September 20, 2010 (identical with TCCA STC SA10-3, Issue 2) | JAR/CS-25 as per EASA.IM.A.191, Issue 4 (incl. SC H-01 "EWIS") The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or (...) | Installation of the Goodrich De-Icer boots must be in accordance with TCCA approved "Goodrich Corporation Installation and Maintenance Instructions, Pneumatic De-Icers for DHC-8 Q 400", Report ER-2010-041, Revision B, dated June 03, 2010. Maintenance of (...) | None | 20.06.2012 |
| 10042937 | GOODRICH CORPORATION | EASA.IM.A.S.01052 | | | B767-300 | Installation of passenger seating arrangement | | Tranco in Master Drawing List 96-025 Revision C, dated April 27, 1996 or later EASA approved revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.07.2005 |
| 10016001 | GOODRICH INTERIORS | EASA.IM.A.S.02161 | REV. 2 | FAA STC ST01691LA | B767-200/-300 | Installation of "Sentry One" Flight Deck Door Camera Surveillance System | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this approval. | Hollingshead International, Inc. Master Data List, Document No. 2765-01, Revision NC, dated October 19, 2004- AFM Supplement, Document No. 2765-AFM, revision NC, 5th November 2004 or later revisions of the above listed documents approved by EASA in accordance with (...) | None | 16.08.2012 |
| 10016462 | GOODRICH INTERIORS | EASA.IM.A.S.02808 | REV. 1 | FAA STC ST01788LA | DC-9-81 (MD-81), DC-9-82 (MD-82), DC-9-83 (MD-83), DC-9-87 (MD-87), MD-88, MD-90-30. | Installation of Sentry One Flight Deck Door Surveillance Camera System. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Hollingshead International Master Data List No. 2788-01 dated 15 March 2005- Flight Manual Supplement Document No. 2788-AFM dated 03 May 2005 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None | 08.11.2012 |
| 10016505 | GOODRICH INTERIORS | EASA.IM.A.S.02869 | REV. 1 | FAA STC ST01007LA-D | 737-800 | Installation of Hollingshead Sentry One Cabin Video Camera System. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Hollingshead International Drawing List 2712-02, Revision A dated 03/06/2003- Flight Manual Supplement Hollingshead International Inc. document No. 2712-AFM, revision NC dated 03/06/2003 or later revisions of the above listed documents approved by EASA (...) | None | 08.11.2012 |
| 10016523 | GOODRICH INTERIORS | EASA.IM.A.S.02896 | REV. 1 | FAA STC ST01638LA | 737-300, 737-400, 737-500 | Installation of Hollingshead Sentry One Cabin mounted Video Camera System. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Hollingshead International Master Data List 2767-01, Revision NC dated 18/05/2004- Flight Manual Supplement Hollingshead International Inc. document No. 2767-AFM, revision Original dated 02/06/2004 or later revisions of the above listed documents approved by (...) | None | 07.11.2012 |
| 10029901 | GOODRICH INTERIORS | | REV. 1 | FAA STC ST01472LA-D | 747-100, 747-100B, 747-100B SUD, 747-200B, 747-300, 747-400, 747-SP | Installation of Hollingshead International Sentry One Cockpit Entry Door Surveillance Camera System. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Hollingshead International Master Data List No. 2763-01, Revision NC dated 25/05/2004- Hollingshead International FAA Flight Manual Supplement, document No. 2763-AFM (->->-) revision NC, dated May 25, 2004- Hollingshead International Flight Manual Supplement (...) | None | 08.11.2012 |
| 10016384 | GOODRICH LIGHTING SYSTEMS, INC | EASA.IM.A.S.02708 | | FAA STC ST02050LA | DC-10-10, DC-10-10A, DC-10-15 DC-10-30F, DC-10-40F, KDC-MD-11 DC-10-40, DC-10-30, DC-10-10F MD-10-10F MD-11-11F, MD-10-30F, | Anticollision Recognition Lighting System (ARLS) in accordance with, Master Drawing List, Issue No. R-366-00 dated 5 December 2007 or later EASA-approved revision. | | Data package containing the following documents, and additional supporting data: 1. FAA ST02050LA2. FAA Letter dated 27th April 20073. Project Specific Certification Plan R-366 Project no. ST11125LA-T.d.t. 12 May 20064. Master Drawing List (...) | This STC is approved only for the product configuration as defined in the approved design data. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 24.04.2008 |
| 10016485 | GOODRICH LIGHTING SYSTEMS, INC | EASA.IM.A.S.02840 | | FAA STC ST10742SC | A318, A319, A320, A321 | Forward LED Navigation Lighting System Installation | | # FAA STC ST10742SC# Aircraft Systems and Manufacturing Top Drawing List, ref. 5685-0001-00, Rev. C dated 19.07.2007# Instruction for Continued Airworthiness, ref. R-398-13, Rev. A, dated 15.11.2007 | | 09.06.2008 |
| 10044480 | GORE DESIGN COMPLETIONS, Ltd. | | REV. 1 | FAA STC ST11099SC | A340-541 | Executive Interior Installation This STC installs a VIP cabin interior including VIP private area with bathroom, bedroom, maill, conference/dining room, hallway, VIP galley/lavatory, executive seating area, staff seating area, medical area and aft galleys. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Conditions: From the original aircraft Type Design : P-1018 : Engine sustained imbalance H-01 : Enhanced Airworthiness (...) | Substantiation Reports List, ref : 538-SD-04 Revision A, dated 23 September 2013 Substantiation Reports List, ref : 538-SD-04-EASA Revision A, dated 12 April 2013 Instruction of Continued Airworthiness, ref : 538-TP-05 revision F, dated 20 September 2013E (...) | The following modifications are prerequisites for this STC installation : * ATD/Aerocon STC, ref : EASA.IM.A.S.01635 (equivalent to FAA STC ST01825LA), acoustic insulation package * ATD/Aerocon STC, ref : 10039327, Revision 2 (equivalent to FAA (...) | 24.09.2013 |
| 10037327 | GPI | | | | AS 332 C, C1, L, L1, L2 | Installation of Cargo Net in Cabin. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Voluntary compliance to the following paragraphs of CS29 Change 2 dated 17 November 200 (...) | SFM-P-10-007 Issue 1 dated 3 November 2011 - Flight Manual Supplement or later revisions of the above listed documents approved by EASA. CMM/P10007 revision 2 dated 3 November 2011 - Cargo Net Component Maintenance Manual; CMX/10-007 issue 4 dated 3 October 2011 - Compliance Matrix. | No Passengers allowed Weight Limitation - 294 kg; Maximum deformation - 257mm | 18.11.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|----------------------|--|--|---------------------|---|--|---------------|
| 10013948 | GRAFAIR AB | EASA.A.S.00421 | REV. 1 | | CESSNA 550 CESSNA S550 | Installation of an Allied Signal Corporation CAS-67A Traffic Alert and Collision Avoidance System (TCAS II). Technical Documentation Associated: - FAA STC ST0011680 - Pro Star Aviation, Master Document List No 0100-MDL-02. Revision A, dated February 1, 2001, or later FAA-approved revision. | | | 1. Pro Star Aviation Aircraft Flight Manual Supplement No 0100-FMS-02 for the Cessna Aircraft Company, Models 550, S550 and 560, Revision A, FAA-approved March 8, 2001, or later FAA-approved revision, is required with this modification.2. This approve (...) | 03.11.2005 |
| 10015541 | GRIMES AEROSPACE COMPANY | EASA.IM.A.S.01425 | | ST500CH | BOEING 737-300 BOEING 737-400 BOEING 737-500 | Installation of Grimes Aerospace Overwing Emergency/ IceDetection Light Assemblies | | - Installation Instructions, Grimes Aerospace Doc. Ref. 30-2471AI, Rev. A dated 10 April 1996- FAA STC ST500CH, dated 22 April 1996 | According to FAA STC ST500CH | 08.11.2005 |
| 10015936 | GROVE AIRCRAFT LANDING GEAR SYSTEMS | EASA.IM.A.S.02077 | | SA01704LA | SEE APPROVED MODEL LIST | Installation of the Grove Disk Brake System in accordance with FAA approved Master Drawing List Number MDL237-00001, Revision B, dated December 8, 2004, or later FAA approved revision are required as part of this installation | | FAA approved Master Drawing List Number MDL237-00001, Revision B, dated December 8, 2004, or later FAA approved revision | The installation applies only to aircraft that have hydraulic brake systems installed. The installation should not be incorporated in any aircraft unless it is determined that the interrelationship between this installation and any previous | 04.07.2006 |
| 10015449 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01192 | | FAA STC ST02883AT-D | GULFSTREAM 200 | Installation of Transponder Flight ID | | -Gulfstream Aerospace Corporation Index List CEB11268000, Revision NC Dated 24 June 2004, or later FAA approved revision-FAA approved Airplane Flight Manual Supplement CEB1126M000/revision NC Dated 13 July 2004 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 28.06.2005 |
| 10015456 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01202 | | FAA STC ST10299SC-D | GULFSTREAM 200 | Installation of ICG-200 SATCOM System | | -Index List GA211038000, Revision B dated 15 February 2005 or later FAA approved revision-FAA approved Airplane Flight Manual Supplement doc. Number GA21103M000 Rev None dated 16February 2005 or later FAA approved revision | 1. Instructions for continued Airworthiness GA24103A000 Revision NC dated 31 January 2005 or later FAA approved revision is required.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the (...) | 28.06.2005 |
| 10015457 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01203 | | FAA STC ST10297SC-D | GULFSTREAM 200 | Installation of Executive Cabin Interior | | -Gulfstream Aerospace Index List GA220008103, Rev A dated 14 February 2005 or later FAA approved revision-FAA approved Flight Manual Supplement GA21468M103, Rev None dated 14 February 2005 or later FAA approved revision | 1. This STC is limited to the installation made in Gulfstream Aerospace LP (Israel Aircraft Industries, LTD) model Gulfstream 200 serial number 103 only.2. This STC does not permit manufacturing of part for multiple installation.3. This STC is approved (...) | 28.06.2005 |
| 10015458 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01204 | | FAA STC ST10298SC-D | DR 140B, DR 180R, DR 200R ROBIN DR 400 RP, DR 120D, | Installation of WX-1000E Stormscope System | | -Gulfstream Index List GA212258000 Revision D dated 17 February 2005 or later FAA approved revision-FAA approved Flight Manual Supplement GA 21225M000 Rev None dated 17 February 2005 or later FAA approved revision | 1. This installation is eligible for airplanes S/N 052 and subsequent with Collins Pro Line 4, version 6.1 avionics system.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph (...) | 28.06.2005 |
| 10015459 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01205 | | FAA STC ST10050SC-D | GULFSTREAM 200 | Installation of Honeywell Laser V IRS | | -Index List GA211138000, Revision B dated 18 August 2004 or later FAA approved revision -FAA approved Airplane Flight Manual Supplement doc. Number GA21113M000 Rev None dated 18 August 2004 or later FAA approved revision | 1. Compliance with FAA Special Condition 25-263-SC and ACJ/EASA CRI F-06 for HIRF has been demonstrated.2. The aircraft must be equipped with Rockwell Collins FMS-6100 Flight Management Systems installed in accordance with IAI MOD Nos.7219 and 100233 (...) | 28.06.2005 |
| 10015460 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01206 | | FAA STC ST10020SC-D | GULFSTREAM 200 | Installation of DeCrane/ERDA Cabin Double Seats | | -Gulfstream Aerospace Index list GA2220080002, Rev D dated 17 May 2003 or later FAA approved revision-FAA approved Flight Manual Supplement GA22208M002 Rev None dated 17 May 2003 or later FAA approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 28.06.2005 |
| 10015579 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01490 | | FAA STC ST09625SC | GULFSTREAM 200 | Installation of a Universal Unilink UL-601 Datalink Data Communication System | | -Master Drawing List Engineering Document Number 30-1100001, Revision K, dated October 2, 2000 or later approved revision- Aeroplane Flight Manual Supplement Rev 1 dated 27 September 2005, or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 06.10.2005 |
| 10015580 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01491 | | FAA STC ST09628SC | GULFSTREAM 200 | Installation of Hard Point Attachments and associated floorstructure | | -Galaxy Aerospace STC data list Engineering Document Number30-1100004, revision E dated 15 November 1999, or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 06.10.2005 |
| 10015581 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01492 | | FAA STC ST09629SC | GULFSTREAM 200 | Installation of an Artex ELT (110-406 NAV) Emergency Locator Transmitter System with NAV interface | | -Engineering Document 30-1100005, Revision K dated June 27, 2000 or later approved revision- Aeroplane Flight Manual Supplement document number 30-112004 dated January 14, 2000 or later approved revision | 1. For aircraft imported into the European Union, changes per Gulfstream Aerospace drawing CEB44231001, Rev A are required to be incorporated until Revision L, on the Engineering document 30-110005 has been released.2. This STC is approved only for t (...) | 06.10.2005 |
| 10015582 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01493 | | FAA STC ST09747SC | GULFSTREAM 200 | Installation of a N°2 Allied Signal KHF-950 HF Communication System | | -Galaxy Aerospace STC Data List number 30-1100009, revisionC, dated April 18, 2000, or later approved revision-Airplane Flight Manual Supplement Revision 1, dated 27 September 2005, or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 06.10.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|-----------------------|---|---------------------|---|--|---------------|
| 10015583 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01494 | | FAA STC ST09748SC | GULFSTREAM 200 | Installation of a second 77 cubic-foot oxygen bottle in nose radome area, above existing bottle | | -Galaxy Aerospace STC data list 30-1100010, revision K, dated June 26, 2003, or later approved revision-FAA Airplane Flight Manual Supplement GA22708M001, revision IR dated July 7, 2003 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 06.10.2005 |
| 10015584 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01495 | | FAA STC ST09851SC | GULFSTREAM 200 | Installation of interior shell, forward cabinets, card tables, lavatory partitions and door, curtain | | -Interior Master Drawing Document number IMDL6600-012, Revision AC, dated August 28, 2002, or later approved revision-Airplane Flight Manual Supplement document number 30-1120014, dated March 12, 2001, or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 06.10.2005 |
| 10015585 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01496 | | FAA STC ST09953SC | GULFSTREAM 200 | Installation of B/E Aerospace, Aircraft Modular Products Forward/Rearward Facing Single Seat, model 2524.546.01 () | | Gulfstream STC Data List 30-1100018, Revision C, dated November 9, 2001, or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 06.10.2005 |
| 10015586 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01497 | | FAA STC ST09954SC | GULFSTREAM 200 | Installation of an Oxygen Filler Panel with Pressure Gauge | | Gulfstream STC Data List 30-1100017, Revision C, dated August 22, 2001, or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 06.10.2005 |
| 10015749 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01782 | REV 2 | FAA STC ST10301SC-D | GULFSTREAM 200 | Installation of Executive Cabin Interior in accordance with Index List GA222048001, Rev A, dated 3-23-2005 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | | Gulfstream Aerospace Index List GA222048001, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision), Approved Flight Manual Supplement GA21204N105, R.(...) | Compatibility of this design change with previously approved modifications must be determined by the installer. | 03.03.2008 |
| 10015750 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01783 | | FAA STC ST10304SC-D | GULFSTREAM 200 | Installation of Forward R/H Galley | | Gulfstream Aerospace Index List GA223288000, Rev. E, dated 4/8/2005, or later FAA approved revision.FAA approved Flight Manual Supplement GA21328M000, Rev. None, dated 4/8/2005 or later FAA approved revision. | Compatibility of this design change with previously approved modifications must be determined by the installer. | 14.03.2006 |
| 10015751 | GULFSTREAM AEROSPACE Corp. | EASA.IM.A.S.01784 | | FAA STC ST02847AT-D | GULFSTREAM 200 | Installation of Safe Flight Instrument Corp Auto Power Automatic Throttle Control System | | Gulfstream Aerospace Index List CEB14348000, Rev. B, dated 6/29/2004, or later FAA approved revision.FAA approved Flight Manual Supplement CEB1434M000, Rev. None, dated 7/2/2004 or later FAA approved revision. | Compatibility of this design change with previously approved modifications must be determined by the installer. | 14.03.2006 |
| 10015382 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01069 | | FAA STC ST01220LA-D | GULFSTREAM GV S/N 641 | Aircraft Serial No. 641 completion and cabin interior installation. | | Installation in accordance with approved Gulfstream Aerospace Corporation Index List GC52 0008277, Revision "D", dated May 13, 2002 or later approved revisions.Operation to be in accordance with Gulfstream Flight Manual Supplement G-7517 AFMS, latest approved revision. | 1. GV Serial No. 641 only | 17.07.2007 |
| 10015388 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01085 | | FAA STC ST01671LA-D | GULFSTREAM GV-SP | Cabin Interior Installation | | -FAA STC ST01671LA-D and associated documentation-GAC Flight Manual Supplement GC551605M006 | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.01.2005 |
| 10015389 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01086 | | FAA STC ST01376LA-D | GULFSTREAM GV-SP | Installation of Aerospace Technologies Group (ATG) Cabin Window | | FAA STC ST01376LA-D and associated documentation | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.01.2005 |
| 10015390 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01087 | | FAA STC ST01373LA-D | GULFSTREAM GV-SP | Installation of Securoplane External Camera System | | -FAA STC ST01373LA-D and associated documentation-Gulfstream Flight Manual Supplement GC51501M007 | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.01.2005 |
| 10015391 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01089 | | FAA STC ST01375LA-D | GULFSTREAM GV-SP | Installation of Airshow CMS Two System | | FAA STC ST01375LA-D and associated documentation | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.01.2005 |
| 10015420 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01150 | | FAA STC ST01752LA-D | GULFSTREAM GV-SP | Gulfstream Aerospace Corporation Modification FAA STC ST 01752 LA-D. Aircraft Completion & Custom Cabin Installation. Gulfstream DAS Report Number GC52000K329 Dated 03/08/05 Flight Manual Supplement ref GC 52000M012 Dated 02/08/05 or later approved revisions | | Gulfstream DAS Report Number GC52000K329 dated 03/08/05 Flight Manual Supplement ref GC 52000M012 dated 02/08/05 or later approved revisions | 1. Serial 5068 only. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.08.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|---|--|---------------------|---|--|---------------|
| 10015452 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01196 | | ST01753LA-D | GULFSTREAM GV-SP | Aircraft completion and cabin interior installation Gulfstream DAS report No. GC52000K341 dated 13 SEP 2005 or later approved revisions. | | Flight Manual Supplement GC52000M014 Aircraft Completion.FAA STC ST01753LA-D | This STC is limited to Gulfstream GV-SP Serial Number 5070 only. | 21.09.2005 |
| 10015487 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01248 | | FAA STC ST03048AT-D | GULFSTREAM GV-SP | Aircraft completion and installation of a custom cabin interior | | In accordance with the FAA STC ST03048AT-D adn Gulfstream Aerospace Corporation Index List No GC520008457 Rev C dated 27 April 2005 or later FAA approve revisions. | GV-SP Aircraft Serial Number 5057 Only.1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 01.07.2005 |
| 10015609 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01545 | | ST02952AT-D | GULFSTREAM GV-SP | Installation of a Satellite communications system to be used for passenger communications. | | Gulfstream DAS Index List ref GC511038022 and report No GC51303K008 dated 8/31/2005 or later approved revisions/FMS GC51103M000FAA STC ref ST02952AT-D. | | 26.09.2005 |
| 10015610 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01546 | | ST01824CH-D | GULFSTREAM 100 GULFSTREAM ASTRA/SPX MD-88 | Installation of the Honeywell MK-V EGPWS, with windshear warning and terrain display on the Collins Pro Line 4 EFIS | | - FAA STC No. ST01824CH-D- Master Drawing List GA113498001, Rev. B, dated March 30, 2005 or later EASA approved Revisions- Airplane Flight Manual Supplement Report No. GA11349M001, Rev. - dated June 24 2003 or later EASA approved Revisions- Instruct (...) | | 12.10.2005 |
| 10015612 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01550 | | FAA STC ST01495LA- | GULFSTREAM GV-SP | Installation of a Honeywell Satellite TV antenna to provide television reception to passenger cabin TV monitors | | -Gulfstream DAS Index List ref GC515070029 and report No GC51507K000 daetd 9/1/2005 or later approved revisions- FAA STCref ST01495LA-D | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 06.10.2005 |
| 10015613 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01551 | | FAA STC ST01378LA-D | GULFSTREAM GV-SP | Installation of a Conduit in the vertical stabiliser to accommodate wiring for the STACOM and SAT TV antennas | | -Gulfstream DAS Index List ref GC524989002 and report No GC52498K000 dated 9/1/2005 or later approved revisions- FAA STCref ST01378LA-D | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 06.10.2005 |
| 10015617 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01569 | | FAA STC ST02915AT-D | GULFSTREAM GV-SP | Lightning Sensor System Installation | | -FAA STC ST02915AT-D-AFMS GC51 225M000 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 06.12.2005 |
| 10015618 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01570 | | FAA STC ST03134AT-D | GULFSTREAM GV-SP | Aircraft Completion STC | | FAA STC Number ST03134AT-D | Limited to S/N 5080 | 06.12.2005 |
| 10015628 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01580 | | FAA STC ST01758LA-D | GULFSTREAM G550 | Installation of Executive Cabin Interior and other Completion Related Modifications | | -FAA STC No ST01758LA-D-GAC Index List GC52 0008491, Rev D,Jan 10, 2006 or later approved revisions-AFMS GC52000M007, Rev 0, Jan 11, 2006 or later approved revisions | G550 S/N 5078 only | 30.01.2006 |
| 10015629 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01581 | | ST01503LA-D | GULFSTREAM G550 | Installation of an Motorola Satellite Telephone (Iridium) Antenna | | -FAA STC ST01503LA-D-GAC Index List GC511038000, Rev D, Dec02, 2003 or later approved revisions | GV-SP (550) | 30.01.2006 |
| 10015673 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01651 | | FAA STC ST03142AT-D | GULFSTREAM GV-SP | Aircraft Completion STC | | FAA STC Number ST03142AT-D | 1. Limited to S/N 50812. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 06.12.2005 |
| 10015694 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01682 | | FAA STC ST03047AT-D | GULFSTREAM GIV-X | Installation of Airshow 4000 (Moving Map and In-Flight Information) | | Gulfstream Aerospace Corporation Index List GC41 5048001, Revision D, dated March 14, 2005 or later approved revisions | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.02.2006 |
| 10015695 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01683 | REV 1 | FAA STC ST03027AT-D | GULFSTREAM GIV-X | Installation of Broad Band Multi-Link System (BBML) | | Gulfstream Aerospace Corporation Index List GC41 3038002, Revision J, dated May 30, 2006 or later approved revision.FAA approved Flight Manual Supplement, GC41 303M001, Revision NC dated May 03, 2005, or later approved revision is required for the-501 (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 19.01.2007 |
| 10015696 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01684 | | FAA STC ST02987AT-D | GULFSTREAM GIV-X | Installation of Interior Emergency Lights | | Gulfstream Aerospace Corporation Index List GC41 4868000, Revision E dated March 14, 2005 or later approved revisions | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.02.2006 |
| 10015697 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01685 | | FAA STC ST02980AT-D | GULFSTREAM GIV-X | Installation of External Camera System (Securaplane) | | Gulfstream Aerospace Corporation Index List GC41 5058000, Revision E, dated March 31, 2005 or later approved revisionsFAA approved Flight Manual Supplement, GC41505M003, Revision NC, dated April 1, 2005, or later approved revisions | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.02.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|--|--|---------------------|---|---|---------------|
| 10015698 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01686 | | FAA STC ST0297AT-D | GULFSTREAM GIV-X | Installation of Passenger Oxygen System | | Gulfstream Aerospace Corporation Index List GC41 4538000, Revision B dated March 07, 2005 or later approved revisions.FAA approved Flight Manual Supplement, GC41453M000, Revision NC, dated March 17, 2005 or later approved revisions | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 08.02.2006 |
| 10015699 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01687 | | FAA STC ST02948AT-D | GULFSTREAM GIV-X | Installation of Cabin Window Shades | | Gulfstream Aerospace Corporation Index List GC41 5068000, Revision C, dated March 14, 2005 or later approved revisions.FAA approved Flight Manual Supplement, GC41506M002, Revision NC, dated March 17, 2005 or later approved revisions | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 08.02.2006 |
| 10015773 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01817 | | FAA STC ST03196AT-D | GULFSTREAM GV-SP | Completion and installation of the custom interior on Aircraft Serial 5094 | | In accordance with Gulfstream Aerospace Corporation Master Index List GC514915094, Revision E, dated March 31, 2006, or later approved revisions.FAA approved Flight Manual Supplement, GC51501M014, Revision NC, dated March 31, 2006, or later approved revisions | Aircraft Serial 5094 Only | 21.04.2006 |
| 10015821 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01893 | | FAA STC ST01519LA-D | GULFSTREAM GV-SP | Completion and installation of the custom interior on Aircraft Serial No 5040 | | In accordance with Gulfstream Aerospace Corporation Master Index List GC520008421, Revision E, dated February 11, 2005, or later approved revisions.FAA approved Flight Manual Supplement, GC51 605M005, Revision NC, dated October 26, 2004, or later approved revisions. | Aircraft Serial 5040 Only | 24.08.2006 |
| 10015839 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.01933 | | FAA STC ST03224AT-D | GULFSTREAM GV-SP | Completion and installation of the custom interior on Aircraft Serial 5099. | | In accordance with Gulfstream Aerospace Corporation Master Index List GC514915099, Revision F, dated May 10, 2006, or later approved revisions.FAA approved Flight Manual Supplement, GC51501M015, Revision N/C, dated May 11, 2006, or later approved revisions. | Aircraft Serial 5099 Only. | 06.06.2006 |
| 10015914 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02044 | | FAA STC ST02920AT-D | GULFSTREAM GV-SP | Installation of Honeywell MCS-7000 SATCOM system | | In accordance with Gulfstream Aerospace Corporation Index List GC51 1038013, Revision B, dated January 17, 2005 or later approved revision.FAA approved Flight Manual Supplement, GC51 103M004, Revision NC, dated January 20, 2005 or later approved revision | None | 21.08.2006 |
| 10015923 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02058 | | FAA STC ST02905AT-D | GULFSTREAM 100 GULFSTREAM ASTRA/SPX | Installation of Collins Transponder with Flight ID for Elementary Surveillance (-501 configuration) or Enhanced Surveillance (-503 configuration). | | 501 config: AFMS CEA1126M000 / IFCA CEA1126A000503 config: AFMS CEA1126M002 / IFCA CEA1126A002 | | 24.07.2006 |
| 10015929 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02065 | | FAA STC ST03254AT-D | GULFSTREAM GV-SP | Completion and installation of the custom interior on Aircraft Serial 5071 | | In accordance with Gulfstream Aerospace Corporation Master Index List GC514915071, Revision D, dated July 3, 2006 or later approvedrevision.FAA approved Flight Manual Supplement, GC51 504M001, Revision NC, dated December 22, 2004 or later approved revision | Aircraft Serial 5071 only | 21.08.2006 |
| 10015930 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02066 | | FAA STC ST03264AT-D | GULFSTREAM GIV-X | Completion and installation of the custom interior on Aircraft Serial Number 4045. | | In accordance with Gulfstream Aerospace Corporation Master Index List GC414914045, Revision D, dated 30 August 2006, or later approved revisions. | Aircraft Serial Number 4045 Only. | 27.10.2006 |
| 10015939 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02081 | | FAA STC ST01673AT-D | GULFSTREAM GV-SP | Installation of a Honeywell P-880 Radar System | | Installed in accordance with Gulfstream Aerospace Corporation Index List GC51 1258001, Revision 02, dated June 20, 1997, or later approved revisions.FAA approved Flight Manual Supplement, GC51 125M001, Revision NC, dated June 30, 1997 or later appro (...) | N/A | 15.03.2007 |
| 10015940 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02082 | | FAA STC ST01673AT-D | GULFSTREAM GV-SP | Installation of a Transponder Flight Identification | | Installed in accordance with Gulfstream Aerospace Corporation Index List CE51 1268000, Revision NC, dated June 24, 2004, or later approved revisions.FAA approved Flight Manual Supplement, CE51 126M000, Revision NC, dated July 12, 2004 or later appro (...) | N/A | 15.03.2007 |
| 10015941 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02083 | | FAA STC ST01388AT-D | GULFSTREAM GV-SP | Installation of a Vacuum Toilet System | | Installed in accordance with Gulfstream Aerospace Corporation Index List GC52 4048002, Revision 02, dated July 02, 1997, or later approved revisions.Continued airworthiness instructions as defined in GC52 404A001, Revision NC, dated July 11, 1997, or later approved revision is required. | N/A | 15.03.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|--------------------------------------|---|--|--|---|---------------|
| 10015969 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02120 | | FAA STC ST03295AT-D | GULFSTREAM GV-SP | Completion and installation of the custom interior on Aircraft Serial 5116 | | In accordance with Gulfstream Aerospace Corporation Master Index List GC51 4915116, Revision D, dated October 20, 2006, or later approved revisions.FAA approved Flight Manual Supplement, GC51 504M001, Revision NC, dated December 22, 2004, or later approved revisions | Aircraft Serial 5116 Only.The following STC's are required to be installed as part of this STC:-FAA STC ST02915AT-D, EASA Approved ref: EASA.IM.A.S.015969,FAA STC ST02796AT-D, EASA Approved ref: EASA.IM.A.S.01259,FAA STC ST02696AT-D, EASA Approved ref (...) | 09.11.2006 |
| 10015988 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02142 | | ST02048CH-D | GULFSTREAM GV-SP | Installation of the Airshow 4000 | | Installation in accordance with Gulfstream Master Drawing List GC515046000, Revision D, dated August 2, 2005, or later approved revisions.Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC51 504M000, Revision #, dated 1 June 2005 | -- | 14.12.2006 |
| 10015989 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02143 | REV. 1 | FAA STC ST02049CH-D | GIV-X, GV-SP | Installation of Securaplane Security System in Gulfstream models GIV-Xand GV-SP | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC514566000, Revision G, dated January 28, 2010, or later approved revisions.or later revisions of the above listed documents approved by EASA in accordance with E (...) | None | 24.02.2011 |
| 10015990 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02144 | | ST02047CH-D | GULFSTREAM GV-SP | Installation of Magnastar Radio Telephone System | | Installation in accordance with Gulfstream Master Drawing List GC511046000, Revision D, dated August 2, 2005, or later approved revisions. | -- | 14.12.2006 |
| 10015993 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02147 | | FAA STC ST02696AT-D | GULFSTREAM GV-SP | Installation of Passenger Oxygen System | | In accordance with Gulfstream Aerospace Corporation Index List GC514538000, Revision B, dated August 5, 2003 or later approved revision.FAA approved Flight Manual Supplement: GC51453M003, Revision NC, dated August 19, 2003 or later approvedrevision i (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 28.08.2006 |
| 10015995 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02151 | | FAA STC ST02290CH-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion Aircraft Serial 5114. | | Install in accordance with Gulfstream Aerospace Report GC522043003, Revision A, and Master Drawing List GC522046037, Revision D, dated November 29, 2006, or later approved revisions.Operate in accordance with Gulfstream Aerospace Flight Manual Suppl (...) | Aircraft Serial No 5114 Only. | 14.12.2006 |
| 10016004 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02167 | REV. 1 | FAA STC ST02796AT-D | GV-SP | Installation of Board Band Multi-Link System (BBML)Revision# the EASA STC EASA.IM.A.S.02167 | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certificated noiseand/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC ST02796AT-D-In accordance with Gulfstream Aerospace Corporation Index ListGC513038002, Revision AH, dated July30, 2010-FAA approved Flight Manual Supplement, GC51303M001, Revision A, datedJuly 24, 2006 (required for the -501 configuration), (...) | None | 20.08.2012 |
| 10016009 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02174 | | FAA STC ST03238AT-D | GULFSTREAM GI-X GULFSTREAM GV-SP | Installation of SATCOM (High Speed Data HD-129). | | In accordance with Gulfstream Aerospace Corporation Index List GC41 3038007, Revision D, dated May 31, 2006, or later approved revisions.FAA approved Flight Manual Supplement, GC41303M007, Revision NC, dated May 31, 2006 or later approved revision i (...) | | 15.05.2007 |
| 10016021 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02187 | | FAA STC ST03256AT-D | GULFSTREAM GIV-X | Installation of an Enhanced Surveillance Flight Identification System with Collins Radio Tuning Units | | Installed in accordance with Gulfstream Aerospace Corporation Index List CE41 126M003, Revision D, dated November 09, 2006, or later approved revision.FAA approved Flight Manual Supplement, CE41 126M006 Revision B, dated November 10, 2006 or later a (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and #Associated Technical Documentation#. Compatibility with other aircraft/engine configurations shall be determined by theinstaller. | 15.03.2007 |
| 10016067 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02246 | | FAA STC ST03324AT-D | GULFSTREAM GV-SP | Completion and installation of the custom interior on Aircraft Serial 5127. | | In accordance with Gulfstream Aerospace Corporation Master Index List GC51 4915127, Revision E, dated January 22, 2007, or later approved revisions.FAA approved Flight Manual Supplement, GC51 501M017, dated January 26, 2007, or later approved revisions | Aircraft Serial 5127 Only.The following STC's are required to be installed as part of this STC:-FAA STC ST02696AT-D, EASA Approved ref EASA.IM.A.S.02147,FAA STC ST03290AT-D, EASA Approved ref EASA.IM.A.S.02255FAA ST02697AT-D was approved prior to EA (...) | 05.02.2007 |
| 10016073 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02255 | | FAA STC ST03290AT-D | GULFSTREAM GIV-X GULFSTREAM GV-SP | Installation of Cabin Window Shades (ATG) | | In accordance with Gulfstream Aerospace Corporation Index List GC51 5068000, Revision A, dated November 16, 2006, or later approvedrevisions.FAA approved Flight Manual Supplement,GC41506M004, Revision NC, dated November 21, 2006, for -401, -403, -40 (...) | None. | 23.01.2007 |
| 10016074 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02256 | | FAA STC ST03351AT-D | GULFSTREAM GIV-X | Airplane Completion on Aircraft Serial Number 4060 | | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC41 4911060, Revision F, dated 05 January 2007 or later approved revision. | Aircraft Serial Number 4060 Only.The following STC's are a required part of this STC:-ST02978AT-D, ST02987AT-D, ST03290AT-D, ST03047AT-D, ST02980AT-D and ST03027AT-D. It is the installer's responsibility to ensure these STC's are all EASA approved. | 23.01.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------------|--------------------------------------|--|---------------------|--|---|---------------|
| 10016087 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02274 | | FAA STC ST64049CH-T | GULFSTREAM GV-SP | Completion and installation of the custom interior on Aircraft Serial 5124 | | In accordance with Gulfstream Aerospace Corporation Master Drawing List GC52 2046051, Revision B, dated January 30, 2007, or later approved revision.FAA approved Flight Manual Supplement, GC52 204M017, dated February 1, 2007, or later approved revisions. | Aircraft Serial 5124 only.The following STCs are required to be installed as part of this STC:FAA STC ST02696AT-D, EASA approved EASA.IM.A.S.02147FAA STC ST02796AT-D, EASA approved EASA.IM.A.S.01259FAA STC ST02047CH-D, EASA approved EASA.IM.A.S.02144 (...) | 19.02.2007 |
| 10016122 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02315 | | | GULFSTREAM GIV-X | Airplane Completion on Aircraft Serial Number 4066. | | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC41 4914066, Revision D, dated 8 March 2007, or later approved revision. | Aircraft Serial Number 4066 Only.The following STC#s are a required part of this STC:-ST02979AT-D, ST02987AT-D, ST03290AT-D, ST03047AT-D, ST02980AT-D and ST03027AT-D. It is the installers responsibility to ensure these STC#s are all EASA approved. | 28.03.2007 |
| 10016141 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02337 | | FAA STC ST02883AT-D | GULFSTREAM 200 | Installation of Transponder Flight ID I.a.w. Index List CEB1 1268000, Rev. NC or later FAA approved version, | | AFMS CEB1 126M000 for -501 and -503 configuration,AFMS CEB1 126M001 for -505 configuration,AFMS CEB1 126M003 for -507 configuration, | | 20.08.2007 |
| 10016142 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02338 | | FAA STC ST03358AT-D | GULFSTREAM 200 | Installation of Universal UNS 1C+ Dual FMS Software upgrade.I.a.w. Index list CEB1 1188000 Rev. B or later approved version. | | AFMS CEB1 118M000 | | 20.08.2007 |
| 10016143 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02339 | | FAA STC ST02872AT-D | GULFSTREAM IV | Installation of Gabies Flight Identification. | | In accordance with Gulfstream Aerospace Corporation Index List CE41 1268004, Revision L, dated June 06, 2007, or later FAA approvedrevision. FAA approved Airplane Flight Manual Supplement, CE41 126M000, Revision NC, dated June 11, 2004, or later FA (...) | Compatibility of this design change with previously approved modifications must be determined by the installer. If theholder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence ofthat permission. | 19.10.2007 |
| 10016144 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02345 | REV 1 | FAA STC ST02228CH-D | GULFSTREAM GIV-X GULFSTREAM GV-SP | Installation of the Heads Up Technologies CMS-400 ChecklistManagement System. | | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514258000, Revision C dated November 28, 2007, or later FAA Approved revision.Operation to be in accordance with Gulfstream Flight Manual Supplement GC414225M000, Re (...) | | 10.01.2008 |
| 10016145 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02346 | | FAA STC ST01931LA-D | GULFSTREAM GV-SP | Aircraft Serial No. 5139 completion and cabin interior installation. | | Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC52 0008556, Revision G, dated 30 May 2007, or later approved revisions.Operation to be in accordance with Gulfstream Flight Manual Supplement GC52000M044, Original rele (...) | GV-SP Serial No. 5139 Only. | 14.06.2007 |
| 10016162 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02385 | | FAA STC ST02348CH-D | GULFSTREAM IV | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 1331. | | Install in accordance with Gulfstream Aerospace Report CE428043002, Revision NC, And Master Drawing List CE428048000, Revision B, dated May 1, 2007, or later approved revisions. Operate in accordance with Gulfstream Flight Manual Supplement CE42 804M00 (...) | Aircraft S/N 1331 Only. | 31.05.2007 |
| 10016187 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02416 | | US FAA STC NO. ST01775LA-D | GULFSTREAM GV-SP | Installation of an Avionics Inc Miniature Quick Access DataRecorder (Mini QAR) | | Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC51 4208000, Revision C, dated 27 September 2006, or later approved revisions. | | 21.06.2007 |
| 10016188 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02417 | | FAA STC ST03409AT-D | GULFSTREAM GIV-X | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4004. | | Install in accordance Custom Engineering Index List, Airplane Custom Interior STC CE425006000 Rev-, dated August 24, 2007, or laterapproved revisions. | Aircraft Serial Number 4004 only.The following STCs are a required part of this STC:-ST02979AT-DST02987AT-D, DST02948AT-DIt is the installers responsibility to ensure these STCs are all EASA approved. | 09.10.2007 |
| 10016201 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02443 | | FAA STC ST01508LA-D | GULFSTREAM GV-SP | Aircraft S/N 5142 completion and cabin interior installation. | | Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC52 2046058, Rev. A, dated 9 July 2007, or later approved revisions.Operations to be in accordancewith Gulfstream Flight Manual Supplement GC52 204M019, original releas (...) | GV-SP S/N 5142 only. | 26.07.2007 |
| 10016205 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02447 | | FAA STC ST01508LA-D | GULFSTREAM GV-SP | Aircraft S/N 5022 completion and cabin interior installation. | | Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC52 0008129, Revision J, dated 16 June2004, or later approved revisions.Operations to be in accordance with Gulfstream Flight Manual Supplement CE1605M000, original rel (...) | GV-SP S/N 5022 only. | 26.07.2007 |
| 10016232 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02496 | | FAA STC SA8223SW-D | GIV, GV | Installation of the Runway Awareness and Advisory System (RAAS) Software for the Honeywell MK V Enhanced Ground Proximity Warning System (EGPWS).Installation of the Runway Awareness and Advisory System (RAAS) Software for the Honeywell MK V Enhanced Ground Proximity Warning System (EGPWS). | | Installation in accordance with Gulfstream Index List CE413498050, revision F dated February 21, 2008 or later FAA approved revision.Operation to be in accordance with Gulfstream Flight Manual Supplements CE41349M008, Original Release, dated Februar (...) | 1. Applies to models G-IV and GV aircraft equipped with Honeywell EGPWS with version -210-210 or later approved software.2. Prior to installation of this modification the installer must determine that the interrelationship between this modifc (...) | 19.07.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|------------------|---|---------------------|--|---|---------------|
| 10016233 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02497 | | FAA STC ST03428AT-D | GULFSTREAM GV-SP | Cabin Interior and Aircraft Completion STC on Aircraft Serial Number 5150. | | # Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915150, Rev. C, dated August 21, 2007 or later approved revision.# To be operated in accordance with FAA approved Airplane Flight Manual Supplement ref. (...) | 1. Aircraft Serial Number 5150 only.2. The following STCs are a required part of this STC: ST02696AT-D, ST02697AT-D and ST03290AT-D. It is the installer's responsibility to ensure these STCs are all EASA approved.3. Prior to installation o (...) | 10.09.2007 |
| 10016234 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02498 | | STC ST03414AT-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5153. | | Installation in accordance with Gulfstream Aerospace Corporation Master Index List, GC514915153 Rev. B, dated September 4, 2007, or later approved revisions. | Aircraft Serial Number 5153 only.The following STCs are a required part of this STC:-ST02696AT-DST02697AT-DST03290AT-DST02715AT-DST02796AT-Dand ST00892LA.It is the installer responsibility to ensure these STCs are all EASA approved. | 09.10.2007 |
| 10016236 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02500 | | FAA STC ST01937LA-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5151. | | Install in accordance with Gulfstream Aerospace Report GC5000K477, Rev. N/C and Index List GC520008576 Rev. K, dated 06September 07, or later approved revisions.Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC52000M045, Re (...) | Aircraft Serial Number 5151 only. | 09.10.2007 |
| 10016245 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02520 | | | GULFSTREAM GIV-X | Aircraft Serial No 4092 completion and cabin interior installation. | | Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC422046119, Revision B, Dated November16, 2007, or later approved revisions.Operation to be in accordance with Gulfstream Flight Manual Supplement GC42204M033, Original (...) | GIV-X Serial No. 4092 Only. | 10.01.2008 |
| 10016282 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02577 | | FAA STC ST03476AT-D | GULFSTREAM GIV-X | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4093. | | Install in accordance with Gulfstream Aerospace Corporation Master Index List, GC14914093 Rev. D, dated November 26, 2007, or later approved revisions. | Aircraft Serial Number 4093 only.The following STCs are a required part of this STC:ST03027AT-DST03047AT-DST02987AT-DST02979AT-DST02987AT-DST03290AT-D | 18.12.2007 |
| 10016289 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02584 | | FAA STC ST0194LA-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5162. | | Install in accordance with Gulfstream Aerospace Report GC52000K511, Rev. N/C and Index List GC520008584, Rev. G, dated 11/30/07, Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC52000M056, revision N/C, dated 11/30/07. | Aircraft Serial Number 5162 only. | 18.12.2007 |
| 10016322 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02631 | | FAA STC ST03533AT-D | GULFSTREAM GV-SP | Installation of AirCell Access II Iridium Satellite Telephone. | | Installation in accordance with Gulfstream Aerospace Corporation Index List GC512038002, Rev. C, dated 2 April 2008, or later FAA approved revision. | Aircraft Serial 5180 and subsequent. | 26.05.2008 |
| 10016323 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02632 | | FAA STC ST03522AT-D | GULFSTREAM GIV-X | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4104. | | Install in accordance with Gulfstream Aerospace Corporation Master Index List, GC414914104 Rev. C, dated February 27, 2008, or later approved revisions in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | 1. Aircraft Serial number 4104 only;2. The following STCs are a required part of this STC:- ST03027AT-D- ST03047AT-D- ST02980AT-D- ST02979AT-D- ST02987AT-D- ST03290AT-D- ST03238AT-D3. It is the installers responsibility to ensure these STCs are all EASA approved.4. P | 26.03.2008 |
| 10016324 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02633 | | FAA STC ST03532AT-D | GULFSTREAM GIV-X | Installation of AirCell Access II Iridium Satellite Telephone. | | Install in accordance with Gulfstream Aerospace Corporation Index List, GC412038004 Rev C, dated April 14, 2008, or later approved revisions. | | 05.05.2008 |
| 10016325 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02634 | | FAA STC ST02519CH-D | GULFSTREAM GV-SP | Aircraft Serial No 5176 completion and cabin interior installation | | Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC52 2046083, Revision A, dated March 24, 2008, or later approved revisions.Operation to be in accordance with Gulfstream Flight Manual Supplement GC52 204M027, Original (...) | GV-SP Serial No. 5176 Only | 13.05.2008 |
| 10016368 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02691 | | FAA STC ST03531AT-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5178. | | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915178, Rev. D, dated 1 May 2008, or later approved revisions. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 26.05.2008 |
| 10016369 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02692 | | FAA STC ST03523AT-D | GULFSTREAM GIV-X | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4103. | | Install in accordance with Gulfstream Aerospace Corporation Master Index List, GC414914103 Rev F, dated March 31, 2008, or later approved revisions. | Aircraft Serial Number 4103 only.The following STCs are a required part of this STC:ST03027AT-D,ST02987AT-D,ST03047AT-D,ST02980AT-D,ST03290AT-Dand ST02979AT-D.It is the installer's responsibility to ensure these STCs are all EASA approved. | 05.05.2008 |
| 10016370 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02693 | | FAA STC ST03524AT-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5174. | | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915174, Rev. E, dated 19 March 2008 or later approved revision.To be operated in accordance with FAA approved Airplane Flight Manual Supplement, ref. GC51501M025, (...) | Aircraft Serial Number 5174 only.The following STCs are a required part of this STC:ST02697AT-D, ST-03290-AT-D, ST02696AT-D.It is the installer's responsibility to ensure that these STCs are all EASA approved. | 07.04.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|-------------------------|------------------|---|---------------------|---|---|---------------|
| 10016371 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02694 | | FAA STC ST03525AT-D | GULFSTREAM GV-SP | Airplane Completion STC on Aircraft Serial Number 5167. | | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915167, Rev C, dated January 7,2008 | Aircraft Serial Number 5167 only.The following STC's are a required part of this STC: ST02696AT-D | 21.01.2008 |
| 10016372 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02695 | | FAA STC ST03526AT-D | GULFSTREAM GIV-X | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4099. | | Install in accordance with Gulfstream Aerospace CorporationMaster Index List, GC414914099 Rev E, dated December 18, 2007, or later approved revisions. | Aircraft Serial Number 4099 only.The following STC's are a required part of this STC:-ST03027AT-DST03047AT-DST02980AT-DST02979AT-DST02987AT-DST03290AT-D.It is the installers responsibility to ensure these STC's are all EASA approved. | 10.01.2008 |
| 10016373 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02696 | | FAA STC ST03479AT-D | GULFSTREAM GV-SP | Airplane Completion STC on Aircraft Serial Number 5166. | | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915166, Rev. C, dated January 10, 2008 or later approved revisions.To be operated in accordance with FAA approved Airplane Flight Manual Supplement, ref GC51501M (...) | Aircraft Serial Number 5166 only.The following STC's are a required part of this STC:ST02696AT-D,ST02697AT-D,ST03290AT-D.It is the installers responsibility to ensure these STC's are all EASA approved. | 24.01.2008 |
| 10016374 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02697 | | FAA STC NO. ST01945LA-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5172. | | Install in accordance with Gulfstream Aerospace Report GC52000K524 Rev. N/C and Index List GC520006597 Rev. J, dated 3 March 2008, or later approved revisions.Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC52000M069 Revision N/C, dated 3 March 2008. | Aircraft Serial Number 5172 only. | 17.03.2008 |
| 10016383 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02707 | | ST02084LA-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5175. | | Install in accordance with Gulfstream Aerospace Report GC52000K534. Rev. N/C and Index List GC520008599, Rev. J, datedApril 18,2008, or later approved revisions.Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC52000M071 Re (...) | Aircraft Serial Number 5175 only. | 13.05.2008 |
| 10016391 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02720 | | FAA STC ST03530AT-D | GULFSTREAM GIV-X | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4106. | | Install in accordance with Gulfstream Aerospace CorporationMaster Index List, GC414914106 Rev. E, dated 26 March 2008,or later approved revisions. | Applicable to Aircraft Serial Number 4106 only.The following STC's are required as part of this STC: FAA STC ST03027AT-D, ST03047AT-D, ST02980AT-D, ST02979AT-D, ST02987AT-D, ST03290AT-D, ST03298AT-D. | 07.04.2008 |
| 10016420 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02756 | | FAA STC ST03543AT-D | GULFSTREAM GIV-X | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4113. | | Installation in accordance with Gulfstream Aerospace Corporation Master Index List, GC414914113, rev. D, dated 19 May 2008, or later approved revisions. | Aircraft Serial Number 4113 only.The following STC's are required part of this STC:ST03027AT-DST02987AT-DST03047AT-DST02980AT-DST03290AT-DST03532AT-DST02979AT-DIt is the installersresponsibility to ensure these STC's are all EASA approved.Prior to ins (...) | 23.05.2008 |
| 10016421 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02757 | | FAA STC ST03410AT-D | GULFSTREAM GIV-X | Installation of a Aft Lavatory Multiple STC | | Install in accordance with Gulfstream Aerospace CorporationIndex List, GC420000478, Rev N, dated January 29, 2009, or later FAA approved revision. | 1. This Aft Lavatory Multiple STC, (FAA ref ST03410AT-D), does not approve the entire airplane completion and interior installationand has the following limitations as defined in document GC414680145, #Guidance for Follow on Installation, Aft Lavato (...) | 08.06.2009 |
| 10016439 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02780 | | FAA STC ST03546AT-D | GULFSTREAM GIV-X | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4115. | | Install in accordance with Gulfstream Aerospace CorporationMaster Index List, GC41 4914115 Rev. D, dated May 22, 2008 or later approved revisions. | Aircraft Serial Number 4115 only.The following STC's are a required part of this STC:ST03027AT-DST02987AT-DST03047AT-DST02980AT-DST03290AT-DST03532AT-DST02228CH-DST02979AT-D | 29.05.2008 |
| 10016440 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02781 | REV. 1 | FAA STC ST03535AT-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5177. | | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915177, Rev. F, dated March 31,2008 or later FAA approved revision. | Aircraft Serial Number 5177 only.The following STC's are a required part of this STC:ST02796AT-T,ST02697AT-D,ST02715AT-D,ST03290AT-D,ST03238AT-Dand ST02696AT-D.It is the installers# responsibility to ensure these STC's are all EASA approved. | 06.10.2010 |
| 10016444 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02790 | | FAA STC ST02085LA-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5179. | | Install in accordance with Gulfstream Aerospace with Gulfstream Aerospace Report GC52000K555, Rev. N/C and Index List GC520008601 Rev. L, dated 1 May 2008, or later approved revisions.Operate in accordance with Gulfstream Aerospace FlightManual Sup (...) | Aircraft Serial Number 5179 only. | 30.05.2008 |
| 10016498 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02860 | | FAA STC ST02523CH-D | GULFSTREAM GV-SP | Aircraft Serial No 5186 completion and cabin interior installation. | | Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC52 2046093, Revision A, dated July 8, 2008* Operation to be in accordance with Gulfstream Flight Manual Supplement GC52 204M032, Original Release, dated July (...) | 1. - GV-SP Serial No. 5186 Only2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon t (...) | 29.07.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|------------------|---|---------------------|---|---|---------------|
| 10016504 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02868 | | FAA STC ST02092LA-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5191. | | Install in accordance with Gulfstream Aerospace Report GC52000K574, Rev. N/C and Index List GC52000B319 Rev. L, dated 25 September 2008, or later approved revisions. Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC52000M078, Revision A, dated 2 September 2008. | 1. Aircraft Serial Number 5191 only. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon t (...) | 11.09.2008 |
| 10016512 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02879 | | FAA STC ST02524CH-D | GULFSTREAM GIV-X | Aircraft completion and cabin interior installation Serial No 4119. | | 1. Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC42 2046129, Revision C, dated July 25, 2008, or later approved revisions. 2. Operation to be in accordance with Gulfstream Flight Manual Supplement GC4 (...) | 1. GIV-X Serial No. 4119 Only. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon (...) | 07.08.2008 |
| 10016513 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02882 | | FAA STC ST02583CH-D | GULFSTREAM GV-SP | Aircraft Serial No 5197 completion and cabin interior installation. | | 1. Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC52 2046103, Revision A, dated September 23, 2008, or later approved revisions. 2. Operation to be in accordance with Gulfstream Flight Manual Supplement GC5 (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 27.10.2008 |
| 10016514 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02884 | | FAA STC ST03570AT | GULFSTREAM 200 | Installation of an Auxiliary Power Unit (APU) Surge Control Valve Modification. | | Install in accordance with Gulfstream Aerospace Corporation Index List, CEB24428000, rev D, dated April 3, 2008, or later approved revisions. Operate in accordance with Civil Airworthiness Authority of Israel (CAA) and FAA Approved Gulfstream 200 F (...) | | 02.07.2008 |
| 10016515 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02886 | | FAA STC ST02578CH-D | GULFSTREAM GV-SP | Aircraft Serial No 5189 completion and cabin interior installation. | | Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC52 2046096, Revision A, dated August 26, 2008, or later approved revisions. Operate in accordance with Gulfstream Flight Manual Supplement GC52 204M031, (...) | 1. GV-SP Serial No. 5189 Only. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the air (...) | 17.09.2008 |
| 10016520 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02892 | | FAA STC ST03565AT-D | GULFSTREAM GIV-X | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4122. | | Install in accordance with Gulfstream Aerospace Corporation Master Index List, GC414914122 Rev. F, dated August 1, 2008 or later approved revisions. | Aircraft Serial Number 4122 only. The following STC's are a required part of this STC: ST03027AT-D- ST02987AT-D- ST03047AT-D- ST02980AT-D- ST03290AT-D- ST03532AT-D- ST03238AT-D- ST02979AT-D- It is the installer's responsibility to ensure these STC's are all EASA approved. | 06.08.2008 |
| 10016521 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02893 | | FAA STC ST03449AT-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5155. | | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915155, Rev. F, dated September 25, 2007 or later approved revision. | Aircraft Serial Number 5155 only. The following STC's are a required part of this STC: ST02796AT-D- ST02697AT-D- ST02715AT-D- ST03290AT-D- ST01775LA-D- ST02696AT-D- ST00892L- In addition, this STC must be installed in conjunction with Cabin Modification STC ST0208 (...) | 05.08.2008 |
| 10016534 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02911 | | FAA STC ST03579AT-D | GULFSTREAM GIV-X | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4123. | | Install in accordance with Gulfstream Aerospace Corporation Master Index List GC414914123 Rev. C, dated July 31, 2008 or later approved revisions. | 1. Aircraft Serial Number 4123 only. 2. The following STC's are a required part of this STC: ST03027AT-D- ST02987AT-D- ST03047AT-D- ST02980AT-D- ST03290AT-D- ST03532AT-D- ST03238AT-D- ST02979AT-D- It is the installer's responsibility to ensure that these STC's are all EASA's | 07.08.2008 |
| 10016539 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02916 | | FAA STC ST02089LA-D | GULFSTREAM GV-SP | Installation of Cabin Interior Modification for Aircraft Serial 5155. | | Install in accordance with Gulfstream Aerospace Report GC52000K569, Rev. N/C dated 29 July 2008 and Index List GC520008627 Rev. D, dated 28 July 2008, or later approved revisions. Operate in accordance with Gulfstream Aerospace Flight Manual Supplement (...) | Aircraft Serial Number 5155 only. | 04.08.2008 |
| 10016547 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02926 | | FAA STC ST02091LA-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5193. | | Install in accordance with Gulfstream Aerospace Report GC52000K581, Rev. N/C and Index List GC520008613 Rev. H, dated 20 August 2008, or later approved revisions. Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC52000M082, Revision N/C, dated 20 August 2008. | Aircraft Serial Number 5193 only. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the a (...) | 02.09.2008 |
| 10016548 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02927 | | FAA STC ST02093LA-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5198. | | Install in accordance with Gulfstream Aerospace Report GC52000K595, Rev. A and Index List GC520008618 Rev. L, dated 11 September 2008, or later approved revisions. Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC52000M084, Revision A, dated 11 September 2008 | 1. Aircraft Serial Number 5198 only. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon t (...) | 30.09.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|-------------------------------------|--|---------------------|---|--|---------------|
| 10016560 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02945 | | FAA STC ST03587AT-D | GULFSTREAM GIV-X | Installation of an Airplane Interior for Aircraft Serial 4130. | | Install in accordance with Gulfstream Aerospace Corporation Master Index List, GC414914130 Rev E, dated October 1, 2008, or later approved revisions. | Aircraft Serial Number 4130 only. The following STCs are a required part of this STC: ST03027AT-DST02987AT-DST03047AT-DST02980AT-DST03290AT-DST03532AT-DST02979AT-D It is the installers responsibility to ensure these STCs are all EASA approved. Prior to (...) | 16.10.2008 |
| 10016561 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02946 | | FAA STC ST03592AT-D | GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5201. | | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915201, Rev. G, dated October 22, 2008 or later approved revision. To be operated in accordance with FAA approved Airplane Flight Manual Supplement, ref GC51501M02 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 04.11.2008 |
| 10016568 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02959 | | FAA STC ST03007AT-D | GV | Installation of Honeywell Integrate Avionics Computers (IAC) with Software Version NZ 6.0. | | Installation of Honeywell Integrated Avionics Computers (IAC) with Software Version NZ 6.0 for -501 and -503 configurations in accordance with Gulfstream Aerospace Corporation Index List CE51 1188001 Rev. D or later approved. Operation in accordance (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 02.10.2008 |
| 10016569 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.02960 | | FAA STC ST03074AT-D | GV | Installation of Honeywell NZ-2000 Third FMS Computer with Software Version NZ 6.0. | | Installation of Honeywell NZ-2000 Third FMS Computer with Software Version NZ 6.0 in accordance with Gulfstream Aerospace Corporation Index List CE51 3088000 Rev C or later approved. Operation to be in accordance with Gulfstream Airplane Flight Manual (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 02.10.2008 |
| 10016596 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.03007 | | FAA STC ST03610AT-D | GULFSTREAM G550 GULFSTREAM GV-SP | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5204. | | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915204, Rev. F, dated November 4, 2008 or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 24.11.2008 |
| 10016597 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.03009 | | FAA STC ST03511AT-D | GULFSTREAM G150 | Installation of Concorde Valve Regulated Lead Acid Main Batteries. | | - Install in accordance with Gulfstream Aerospace Corporation Index List, CEC1 4448000 Rev C, dated December 13, 2007, or later approved revisions. - Operation in accordance with Gulfstream Aerospace Corporation Airplane Flight Manual Supplement (...) | None. | 18.02.2009 |
| 10016602 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.03016 | | FAA STC ST02097LA-D | GULFSTREAM G550 | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5203. | | Install in accordance with Gulfstream Aerospace Report GC52000K607 Rev. - , and Index List GC520008622 Rev. J, dated 17 Nov 08, or later approved revisions. Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC52000M086 Revision - , dated 17 Nov 08. | 1. Aircraft Serial Number 5203 only. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect (...) | 02.12.2008 |
| 10016607 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.03024 | | FAA STC ST03617AT-D | GULFSTREAM GIV-X | Installation of Export Modifications for Aircraft Serial 4091. | | Install in accordance with Gulfstream Aerospace Corporation Export Modifications Index List, GC420008128, Rev C, dated October 7, 2008, or later approved revisions. | 1. Aircraft Serial Number 4091 only. 2. This STC must be installed in conjunction with Airplane Completion STC ST03477AT-D (EASA.IM.A.S.03025) to fully meet EASA requirements. 3. It is the installers responsibility to ensure these STCs are all EASA approved (...) | 15.10.2008 |
| 10016608 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.03025 | | FAA STC ST03477AT-D | GULFSTREAM GIV-X | Airplane Completion and Custom Cabin Installation for Aircraft Serial 4091. | | Install in accordance with Gulfstream Aerospace Corporation Master Index List, GC414914091 Rev C, dated October 30, 2007, or later approved revisions. | Aircraft Serial Number 4091 only. The following STCs are a required part of this STC: ST03027AT-DST02987AT-DST03047AT-DST02980AT-DST03290AT-DST02979AT-DST00892LA Additionally, this STC must be installed in conjunction with Export Modification STC ST0361 (...) | 15.10.2008 |
| 10016613 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.03038 | | FAA STC ST03643AT-D | GULFSTREAM GV-SP | Aircraft completion and Installation of Airplane Interior for Aircraft Serial 5210. | | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915210, Rev. G, dated December 10, 2008 or later FAA approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 08.01.2009 |
| 10016621 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.03052 | | FAA STC ST02632CH-D | GULFSTREAM GV-SP | Aircraft completion and cabin interior installation for Serial No 5212. | | Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC52 2046121, Revision B, dated February 5, 2009, or later approved revisions. Operation to be in accordance with Gulfstream Flight Manual Supplement GC52 204M038 (...) | 1. Serial No. 5212 Only. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 02.03.2009 |
| 10016639 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.03079 | | FAA STC ST6870CH-T | GULFSTREAM GV-SP | Aircraft Serial No 5220 completion and cabin interior installation. | | Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC52 2046129, Revision A, dated March 26, 2009, or later approved revisions. Operation to be in accordance with Gulfstream Flight Manual Supplement GC52 204M040, (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 08.04.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|---------------------|----------------|----------------------|------------------|---|---|--|--|---------------|
| 10016650 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.03095 | | FAA STC ST03542AT-D | GULFSTREAM GIV-X | Installation of an Aft Galley/Galley Annex Multiple STC. | | Install in accordance with Gulfstream Aerospace Corporation Index List, GC415408000 Rev L, dated February 04, 2009, or later FAA approved revision. | 1. This Aft Galley/Galley Annex Multiple STC, (FAA ref ST03542AT-D), does not approve the entire airplane interior installation and has the following limitations as defined in document GC415400286: #Guidance for Follow on Installation, Aft Galley/Gall (...) | 10.06.2009 |
| 10016666 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.03139 | | FAA STC ST03668AT-D | GULFSTREAM GIV-X | Installation of an Airplane Interior and completion for Aircraft Serial 4148. | | Install in accordance with Gulfstream Aerospace Corporation Master Index List, GC414914148 Rev E, dated March 17, 2009, or later FAA approved revision. To be operated in accordance with FAA approved Airplane Flight Manual Supplement, ref. GC41204M002, Rev. NC dated 03/17/09, or later FAA approved rev. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 07.04.2009 |
| 10016678 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.03201 | REV. 1 | FAA STC ST03646AT-D | GIV-X | Installation of Cabin/Baggage Compartment | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Install in accordance with Gulfstream Aerospace Corporation Index List GC420004142<->-> Rev. D, dated January 30, 2009-Operation in accordance with Airplane Flight Manual Supplement, GC41204M000, Rev. NC dated January 30, 2009 or later revisions of (...) | This Cabin/Baggage Compartment Multiple STC, (FAA ref ST03646AT-D), does not approve the entire airplane interior installation and has the following limitations as defined in document GC422040204, "Guidance for Follow on Installation, Cabin/Baggage Com (...) | 08.02.2012 |
| 10016679 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.03202 | REV. 1 | FAA STC ST03648AT-D | GIV-X | Installation of Cabin/Baggage Compartment (AG) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Install in accordance with Gulfstream Aerospace Corporation Index List GC420004144<->-> Rev. L, dated 26 February 2010-Operation in accordance with Airplane Flight Manual Supplement, GC41204M001, Rev. NC dated 4 February 2010 or later revisions of t (...) | This Cabin/Baggage Compartment Multiple STC, (FAA ref ST03648AT-D), does not approve the entire airplane interior installation and has the following limitations as defined in document GC422040205, "Guidance for Follow on Installation, Cabin/Baggage Com (...) | 08.02.2012 |
| 10016680 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.03203 | | FAA STC ST03647AT-D | GIV-X | Installation of a Forward Galley-Vestibule Area-Forward Lavatory-Flight Deck Multiple STC | | Install in accordance with Gulfstream Aerospace Corporation Index List, GC421048004 Rev D, dated January 30, 2009, or later FAA approved revision. | 1. The Forward Galley-Vestibule Area-Forward Lavatory-Flight Deck Multiple STC, (FAA ref ST03647AT-D), does not approve the entire airplane completion and interior installation and has the following limitations as defined in document GC421040019, #Gu (...) | 08.06.2009 |
| 10016681 | GULFSTREAM AEROSPACE CORPORATION | EASA.IM.A.S.03204 | | FAA STC ST03649AT-D | GULFSTREAM GIV-X | Installation of a Flight Deck-Forward Vestibule-Forward Lavatory Multiple STC | | Install in accordance with Gulfstream Aerospace Corporation Index List, GC421048003 Rev E, dated February 04, 2009, or later FAA approved revision. | 1. This Flight Deck-Forward Vestibule-Forward Lavatory Multiple STC, (FAA ref ST03649AT-D), does not approve the entire airplane completion and interior installation and has the following limitations as defined in document GC421040020#Guidance for Fo (...) | 10.06.2009 |
| 10027065 | GULFSTREAM AEROSPACE CORPORATION | P-EASA.IM.A.S.03148 | | | GV-SP (G550) | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5224. | JAR 25 | Install in accordance with Gulfstream Aerospace Report GC52000K634, Rev.N/C, dated 11 June 2009 and Index List GC520008647 Rev. G, dated 06-08-09, or later approved revisions Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC5 (...) | Aircraft Serial Number 5224 only. | 03.09.2009 |
| 10027320 | GULFSTREAM AEROSPACE CORPORATION | P-EASA.IM.A.S.03251 | | FAA STC ST03710AT-D | GULFSTREAM GIV-X | Installation of an Airplane Interior and completion for Aircraft Serial Number 4161. | JAR 25. The Certification Basis for the original product remains applicable to this certificate. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate. | Installed in accordance with Gulfstream Aerospace Corporation Index List, GC420004161, Rev D, dated August 25, 2009. To be operated in accordance with Airplane Flight Manual Supplement GC41204M006, Rev NC, date August 25, 2009 or later revisions of the (...) | Aircraft Serial Number 4161 only. The following STCs are a required part of this STC: ST02987AT-D, ST03290AT-D, ST02979AT-D, ST03410AT-D and ST03647AT-D. It is the installer's responsibility to check these STC#s are EASA Approved. | 21.09.2009 |
| 10027731 | GULFSTREAM AEROSPACE CORPORATION | P-EASA.IM.A.S.03254 | | FAA STC ST02705CH-D | GV-SP | Aircraft Serial No 5240 completion and cabin interior installation. | JAR 25 The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC52 2046146, Revision B, dated October 6, 2009, or later approved revisions. Operation to be in accordance with Gulfstream Flight Manual Supplement GC52 2046M04 (...) | GV-SP Serial No. 5240 Only. | 26.10.2009 |
| 10028267 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST03740AT-D | GV-SP | Aircraft Serial No 5246 completion and cabin interior installation. | JAR 25 The Certification Basis for the original product remains applicable to this certificate/ approval | Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC52 2046159, Revision B, dated November 30, 2009, or later approved revisions. Operation to be in accordance with Gulfstream Flight Manual Supplement GC52 2046M047, (...) | GV-SP Serial No. 5246 Only. | 14.12.2009 |
| 10028488 | GULFSTREAM AEROSPACE CORPORATION | P-EASA.IM.A.S.02713 | REV. 1 | FAA STC ST03699AT | GULFSTREAM 150 | G150 Steep Approach Landing | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval See CRI D-06 Special Condition "Steep Approach Landing" | FAA Approved Airplane Flight Manual Steep Approach Supplement 2009-01 for Gulfstream Model G150, Rev 1, FAA approved 4 Sept 09, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 27.01.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|----------------------|----------------|-------------------------|--------------------------|--|---|--|---|---------------|
| 10028505 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC NO. ST03746AT-D | GV-SP | Completion and Installation of an Airplane Interior for Aircraft Serial5242. | JAR 25 | Install in accordance with Gulfstream Aerospace Corporation Master Index List, GC514915242 Rev C, dated December 17, 2009, or later approved revisions. To be operated in accordance with FAA approved Airplane Flight Manual Supplement, ref. GC51468M008, Re (...) | Aircraft Serial Number 5242 only. | 18.01.2010 |
| 10028642 | GULFSTREAM AEROSPACE CORPORATION | | REV. 1 | FAA STC ST03739AT-D | G150 | Installation of an International Communications Group (ICG). Inc.NxtMail Server | JAR 25, Change 15The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Index List GA314288000, Rev. B, dated 05. November 2009-AFMS GA31428M001, Initial Release, dated 05. November 2009-Instructions for continued airworthiness GA34428A000, Initial Release, dated 05 October 2009or later revisions of the above listed doc (...) | Installation of Cabin Interior STC EASA.IM.A.S.02543 (FAA STC ST10316SC-D) is required prior to installation of this NxtMail Server arrangement. | 01.02.2010 |
| 10029061 | GULFSTREAM AEROSPACE CORPORATION | | REV. 1 | FAA STC ST03752AT-D | GIV-X | Aircraft completion and installation of an Airplane Interior, S/N 4180 | JAR 25The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Install in accordance with Master Index List GC414914180, Revision D, dated 10 February, 2010, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | This approval is limited to S/N 4180 only. The following STC's are required part of this STC: ST02979AT-D, ST02980AT-D, ST02987AT-D, ST03027AT-D, ST03047AT-D, ST03290AT-D, ST03410AT-D, ST03532AT-D, and ST03647AT-D. It is the installers responsibility to ensure all these STC's are EASA approved. | 11.06.2010 |
| 10029212 | GULFSTREAM AEROSPACE CORPORATION | | REV. 1 | FAA STC ST03738AT-D | GIV-X, GV-SP | Installation of Fwd & Aft Lavatory Smoke Detectors. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installed in accordance with Gulfstream Aerospace Corporation Index List, ref. GC414558002, Revision G, dated 13th September 2010-Instruction for Continued Airworthiness, ref. GC42455A000, Revision A, dated 28th July 2010 or later revisions of the above (...) | None | 10.05.2012 |
| 10029213 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST03667AT-D | GULFSTREAM GV-SP, GIV-X | Installation of a Generation 3, Broad Band Multi-Link (BBML) System. | JAR 25. The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Install in accordance with Gulfstream Aerospace Corporation Index List GC413038009, Revision E, dated 10 September, 2009. Aircraft to be operated in accordance with: Airplane Flight Manual Supplement, GC41303M012, Revision NC, dated September 10, 2009, or Airplane Flight Manual Supplement, GC41303M013.S | The installer must determine whether this design change is compatible with previously approved modifications. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. | 12.03.2010 |
| 10030647 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST03731AT-D | GIV-X | Installation of an Audio Video Distribution System | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Installation in accordance with Gulfstream Index List GC415028001, Revision D, dated 10 September 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None. | 29.06.2010 |
| 10030690 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC REF ST03737AT-D | GIV-X AND GV-SP | Installation of a Honeywell/EMS HD-710 combined SATCOM Swift Broadband/High Speed Data System. | JAR 25The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Install in accordance with Gulfstream Aerospace Index List ref GC513038009, Revision B, dated 10/09/09 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | Not Applicable | 02.07.2010 |
| 10030691 | GULFSTREAM AEROSPACE CORPORATION | P-EASA, IM.A.S.03225 | | FAA STC REF ST03685AT-D | GV-SP (G550) | Installation of Flight Crew Sleeping Facilities. | JAR 25The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Install in accordance with Gulfstream Aerospace Index List ref GC515018000, Revision E, dated 06/18/09 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | Not Applicable | 02.07.2010 |
| 10031140 | GULFSTREAM AEROSPACE CORPORATION | EASA, IM.A.S.02821 | | FAA STC ST03713-AT | G150 | Installation of Enhanced Vision System (EVS) II | The Certification Basis for the original product remains applicable to this certificate/ approval. JAR 25 Change 15 (referto TCDS)The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | G150 EVS II Master Program Drawing List, G150-GER-0039, dated 26. March 2009EVS II Flight Manual Supplement 2009-02, Revision 1, dated 05 May 2010Instructions for Continued Airworthiness, G150-GER-0037, Rev. -- Dated 05. Jan. 2009 or later revisions of (...) | This approval does not constitute any operation credit | 28.07.2010 |
| 10031982 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST02525CH-D | GULFSTREAM GV-SP (G 550) | Installation of an Aft Lavatory Compartment. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Gulfstream Aerospace Index List GC524048003, Rev F, dated May 6, 2010, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | none | 29.09.2010 |
| 10031983 | GULFSTREAM AEROSPACE CORPORATION | | REV. 1 | FAA STC ST03743AT-D | GV-SP | Installation of an 18 Passenger Aircraft Cabin Compartment. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Installation in accordance with Gulfstream Aerospace Corporation Index List GC52008667, Rev F, dated November 24, 2009. Operation in accordance with Gulfstream Aerospace Corporation Aircraft Flight Manual Supplement ref GC5200M112, Original Release, d (...) | None | 23.02.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|--------------|----------------|-------------------------|--------------------------|--|--|--|--|---------------|
| 10031985 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST02190LA-D | GULFSTREAM GV-SP (G 550) | Installation of the Forward Vestibule Compartment. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Gulfstream Aerospace Corporation Index List GC520008140, Revision F, dated July 2, 2009. Operation in accordance with Gulfstream Aerospace Corporation Aircraft Flight Manual Supplement ref GC52484M001, Original Release, d (...) | none | 29.09.2010 |
| 10032074 | GULFSTREAM AEROSPACE CORPORATION | | REV. 1 | FAA STC ST04204AT-D | GV-SP | Installation of the Aircraft Cabin Compartment for Gulfstream GV-SP Serial Number 5284. | The Certification Basis for the original product remains applicable to this certificate/ approval. JAR 25 Amd 15 The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC520008702 Rev F, dated September 27, 2010. Operation in accordance with Gulfstream Aerospace Corporation Aircraft Flight Manual Supplement No. GC52000M145 Rev -, dated Se (...) | Aircraft Serial Number 5284 Only. | 23.02.2012 |
| 10032075 | GULFSTREAM AEROSPACE CORPORATION | | REV. 1 | FAA STC NO. ST03793AT-D | GULFSTREAM GV-SP | Installation of an Airplane Interior for aircraft S/N 5277 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Install in accordance with Master Index List GC514915277, Revision C, dated August 27, 2010. To be operated in accordance with FAA approved Airplane Flight Manual Supplement, GC51468M014, Rev "-", dated August 26, 2010, or later revisions of the above li (...) | This approval is limited to only the installation made in Gulfstream, GV-SP, A/C 5277. The following STC#s are a required part of this STC: ST02696AT-D, ST02697AT-D, ST03533AT-D, ST02796AT-D, ST02715AT-D, and ST03290AT-D. It is the installers responsibility to ensure these STC's are EASA approved. | 06.10.2010 |
| 10032076 | GULFSTREAM AEROSPACE CORPORATION | | REV. 1 | FAA STC ST02292CH-D | GV-SP | Installation of the 115V AC/60Hz Outlets Multiple STC for Gulfstream GV-SP model aircraft. | The Certification Basis for the original product remains applicable to this certificate/ approval. JAR 25 Amd 15 The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC514948000 Rev D, dated May 30, 2008. Operation in accordance with Gulfstream Aerospace Corporation Aircraft Flight Manual Supplement No. GC41494M000 Rev -, dated November (...) | none | 23.02.2012 |
| 10032077 | GULFSTREAM AEROSPACE CORPORATION | | REV. 1 | FAA STC ST03794AT-D | GV-SP | Installation of the Aircraft Cabin Compartment for Gulfstream GV-SP model aircraft. | The Certification Basis for the original product remains applicable to this certificate/ approval. JAR 25 Amd 15 The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC520008695 Rev E, dated September 3, 2010. Operation in accordance with Gulfstream Aerospace Corporation Aircraft Flight Manual Supplement No. GC52000M137 Rev -, dated Sept (...) | None | 23.02.2012 |
| 10032086 | GULFSTREAM AEROSPACE CORPORATION | | REV. 1 | FAA STC ST03759AT-D | GV-SP | Installation of the Audio Video Distribution System (AVDS) Multiple STC for Gulfstream GV-SP model aircraft. | The Certification Basis for the original product remains applicable to this certificate/ approval. JAR 25 Amd 15 The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC525128001 Rev B, dated March 19, 2010. Operation in accordance with Gulfstream Aerospace Corporation Aircraft Flight Manual Supplement No. GC51512M001 Rev -, dated March (...) | None | 23.02.2012 |
| 10032609 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST04209AT-D | GIV-X | Aircraft Completion and Installation of an Airplane Interior | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC4149144200, Revision E, dated October 20, 2010. Instructions for Continued Airworthiness Deviations Supplement GC42801A432, Revision A, dated October 19, 2010, or later (...) | You may not use descriptive data about this design change to duplicate other products. This approval is limited to aircraft Serial 4200. This STC does not permit manufacturing of parts for multiple installations. | 17.11.2010 |
| 10032631 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST04213AT-D | GIV-X | Aircraft Completion and Installation of Airplane Interior. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC414914202, Revision J, dated November 06, 2010. Instructions for Continued Airworthiness Deviations Supplement GC42801A443, Revision A, dated November 05, 2010, or lat (...) | You cannot use descriptive data about this design change to duplicate other products. This approval is limited to Gulfstream GIV-X, Serial 4202. This STC does not permit manufacturing of parts for multiple installations. | 18.11.2010 |
| 10032651 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST03742AT-D | GIV-X | Installation of a Cabin/Baggage Compartment. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Installation in accordance with Gulfstream Aerospace Corporation Index List GC420004171, Revision G, dated November 02, 2010. Operation in accordance with Airplane Flight Manual Supplement GC41204M009, Revision NC, dated November 20, 2009, or later revl (...) | None | 19.11.2010 |
| 10032980 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST04215AT-D | GIV-X | Aircraft Completion and Installation of an Airplane Interior on Aircraft Serial 4203. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC414914203, Revision F, dated November 09, 2010. Aircraft to be maintained in accordance with Instructions for Continued Airworthiness Deviations Supplement GC42801A431 (...) | You cannot use descriptive data about this design change to duplicate other products. This approval is limited to only the installation made in Gulfstream GIV-X, A/C 4203. This STC does not permit manufacturing of parts for multiple installations. | 13.12.2010 |
| 10033988 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST04225AT-D | GV-SP | Aircraft Serial No 5302 completion and cabin interior installation | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC522046210, Revision A, dated January 24, 2011, or later approved revisions. Operation to be in accordance with Gulfstream Flight Manual Supplement GC52204M061, Ori (...) | GV-SP Serial No. 5302 Only | 24.02.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|--------------|----------------|-------------------------|------------------|--|---|--|---|---------------|
| 10033989 | GULFSTREAM AEROSPACE CORPORATION | | REV. 1 | FAA STC ST04221AT-D | GV-SP | Aircraft Serial No 5294 completion and cabin interior installation | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC522046203, Revision A, dated December 7, 2010, or later approved revisions. Operation to be in accordance with Gulfstream Flight Manual Supplement GC52204M059, Ori (...) | GV-SP Serial No. 5294 Only | 04.03.2011 |
| 10035124 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST03070AT | GIV-X | Installation of a Honeywell LSZ-860 Lighting Sensor System. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Gulfstream Aerospace Corporation Index List GC412258000, Revision F, dated March 28, 2005. Operation in accordance with Gulfstream Flight Manual supplement ref GC41225M000 dated 05 April 2005. "Honeywell LSZ-860 Lighting (...) | This approval should not be extended to other aircraft of this model on which other previously approved modifications are incorporated, unless it is determined by the installer that the interrelationship between this change and any other previously approved... | 27.05.2011 |
| 10035125 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST03566AT-D | GIV-X | Installation of a Custom Cabin Interior and completion of Aircraft Serial Number 4121. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Gulfstream Aerospace Corporation Index List GC420004121, Revision E, dated August 21, 2008 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | You cannot use descriptive data about this design change to duplicate other products. This approval is limited to only the installation made in Gulfstream GIV-X, A/C 4121. This STC does not permit manufacturing of parts for multiple installations. | 27.05.2011 |
| 10035279 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST03795AT-D | GV-SP | Aircraft Completion and Installation of an Airplane Interior | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation of an Airplane Interior in accordance with Gulfstream Aerospace Corporation Index List GC520005280, Revision D, dated September 23, 2010. Operation in accordance with Gulfstream Aircraft Flight Manual Supplement ref GC51468M015, dated 23 S (...) | The installer must determine whether this design change is compatible with previously approved modifications. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. | 10.06.2011 |
| 10035421 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC NO. ST04238AT-D | GV-SP (G550) | Installation of Cabin Interior and Aircraft Completion for Gulfstream GV-SP model Aircraft | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC520008724 Rev G, dated June 8, 2011. Operation in accordance with Gulfstream Aerospace Corporation Aircraft Flight Manual Supplement No. GC52000M149 Rev -, dated June 8, (...) | None | 21.06.2011 |
| 10037106 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST04244AT-D | GULFSTREAM GV-SP | Executive Aircraft Completion, Multiple STC. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Gulfstream Aerospace Corporation Index List GC522048010, Revision -, dated October 18, 2011. Operation to be in accordance with Gulfstream Flight Manual Supplement GC52204M068, Original Release, dated October 18, 2011. Th (...) | None | 03.11.2011 |
| 10038524 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST03784AT-D | GV-SP | Installation of an Audio Video Distribution System | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Gulfstream Aerospace Corporation Index List GC515128003, Revision C, dated June 23, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | 1. The installer must determine whether this design change is compatible with previously approved modifications. 2. If the holder agrees to permit another person to alter the product, the holder must give the other person written evidence of that permission. | 23.02.2012 |
| 10038629 | GULFSTREAM AEROSPACE CORPORATION | | REV. 2 | FAA STC ST04251AT-D | GULFSTREAM GV-SP | Installation of an Executive Completion - Multiple STC | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Gulfstream Aerospace Corporation Index List GC522048013, Revision A, dated 15th February 2012. Operation to be in accordance with Gulfstream Flight Manual Supplement GC52204M075, Rev. A, dated 15th February 2012 or later (...) | This Executive Completion, Multiple STC, (FAA ref ST04251AT-D), does not approve the entire airplane interior installation and has the following limitations as defined in document GC522043046, "Guidance for Follow-On Installation, Executive Completion (...) | 23.03.2012 |
| 10038724 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST03686AT-D | GV-SP | Installation of an Airplane Cabin Interior. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation of an Airplane Interior in accordance with Gulfstream Aerospace Corporation Index List GC520009657, Revision AC, dated March 02, 2012, or later FAA approved revision. Operation in accordance with Airplane Flight Manual Supplement (>-<) >GC5 (...) | 1. The approved EASA Configurations and the compatible EASA export option are defined in GC522040616, "EASA Validated Configuration Lists" (<->-> Revision "...", dated 03/02/12 or later approved revisions. 2. Follow-On Installation Guide GC522040665, Rev (...) | 09.03.2012 |
| 10040608 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST03779AT-D | GIV-X | Installation of an AirCell High Speed Internet System | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC ST03779AT-D, issued 4th June 2010. Installation in accordance with Gulfstream Aerospace Corporation Index List GC413028000, Revision B, dated 4th June 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA (...) | EASA STC number EASA.I.M.A.S.02633 is required for installation of this STC | 13.07.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|--------------|----------------|----------------------|--------------|---|---|---|--|---------------|
| 10040627 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST00803LA-D | GV | Installation of Satcom Antenna (Racal). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - FAA STC ST00803LA-D, issued 1st September 1999;- Installed in accordance with Gulfstream Aerospace Corporation IndexList, ref. IL11011575, Revision C, dated 31st August 1999, or later revisions of the above listed documents approved by EASA in accordance with (...) | None. | 16.07.2012 |
| 10041062 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST03771AT-D | GV-SP | Installation of an Airplane Interior | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.-CRI D-50 "Seat deformation into the overwing Emergency Exits and Emergency Egress." - (...) | -FAA STC ST03771AT-D issued 5th May 2010-Installation in accordance with Gulfstream Aerospace Corporation IndexList GC520005266, Revision C, dated May 5, 2010-Operation to be in accordance with Gulfstream Flight Manual Supplement GC51468M012, Rev. NC (...) | Limitations are defined in document GC522040593, "Follow-On Installation Guide", Revision NC, dated May 5, 20101. The installer must determine whether this design change is compatible with previously approved modifications.2. If the holder agrees to pe (...) | 20.08.2012 |
| 10041063 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST02884AT-D | GV | Installation of a Transponder Flight Identification | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -FAA STC ST02884AT-D, amended 21st October 2009-Installed in accordance with Gulfstream Aerospace Corporation IndexList CE511268000, Revision J, dated October 21, 2009-Airplane Flight Manual Supplement CE51126M000, Revision NC, dated July 12, 2004-A (...) | None | 20.08.2012 |
| 10041101 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST04263AT-D | GV-SP | Executive Completion, Multiple STC. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.-CRI D-50 "Seat deformation into the overwing Emergency Exits and Emergency Egress." - (...) | -, FAA STC ST04263AT-D issued 14th August 2012;- Installation in accordance with Gulfstream Aerospace Corporation IndexList GC522048017, Revision A, dated August 14, 2012;- Operation to be in accordance with Gulfstream Flight Manual Supplement GC522 (...) | Limitations are defined in document GC522043055, "Follow-On Installation Guide", Revision NC, dated August 7, 20121. The installer must determine whether this design change is compatible with previously approved modifications.2. If the holder agrees (...) | 21.08.2012 |
| 10041234 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST01105LA-D | GV | Installation of a 13 Passenger Cabin Interior | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - FAA STC ST01105LA-D, issued 26th June 2001- Installed in accordance with Gulfstream Aerospace Corporation IndexList ref. GC520008262, Revision A, dated 25th June 2001 or later revisions of the above listed documents approved by EASA in accordance with (...) | The approval of this modification applied only to Model GV S/N 622 | 31.08.2012 |
| 10041649 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST01822AT-D | GV AND GV-SP | Installation of APU Fire Bottle | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installed in accordance with Gulfstream Aerospace Corporation IndexList GC514428001, Revision L, dated July 31, 2011. Operation in accordance with AFM GC51442M001, dated April 8, 1999 or later FAA approved revision required for fully operational config (...) | 1- The installer must determine whether this design change is compatible with previously approved modifications.2- If the holder agrees to permit another person to alter the product, the holder must give the other person written evidence of that permission. | 04.10.2012 |
| 10041900 | GULFSTREAM AEROSPACE CORPORATION | | REV. 1 | FAA STC ST00673LA-D | GV | Installation of Honeywell TCAS 2000 TCAS II System | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -FAA STC ST00673LA-D(->-<-) issued 17th May 1999-Installed in accordance with Gulfstream Aerospace Corporation IndexList ref. GC511298001, Revision C, dated 26th April 1999-Airplane Flight Manual Supplement, ref. GC51129M001, Rev. A, dated 26th Feb (...) | None | 23.10.2012 |
| 10042374 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST01751LA-D | GV-SP | Installation of EROS Convertible MLD20 Series Crew Oxygen Mask -Regulator with Detachable Smoke Goggles | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -FAA STC ST01751LA-D, amended 13th June 2011-Installed in accordance with Gulfstream Aerospace Corporation IndexList, ref. GC544538000, Revision E, dated 8th June 2011-EASA/JAR Compliance Report, ref. CE528043005, Revision -, dated 12th November 2012 (...) | None | 28.11.2012 |
| 10042377 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST01957CH-D | GV, GV-SP | Installation of an Airshw RF/IF Remote Controller | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -FAA STC ST01957CH-D, re-issued 31st July 2012-Installed in accordance with Gulfstream Aerospace Corporation Master Drawing List GC516028000, Rev. A, dated 8th September 2004.-EASA/JAR Compliance Report, ref. CE528043004, Revision -, dated 7th November (...) | None | 28.11.2012 |
| 10042407 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST02518CH-D | GV-SP | Installation of a 16 Passenger Corporate Aircraft Interior | JAR 25, Change 15The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are (...) | -FAA STC ST02518CH-D, re-issued 9th October 2012-Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List GC522048078, Revision A, dated March 13, 2008-EASA/JAR Compliance Report, ref. CE528043003, Revision -, dated 16th Nov (...) | This STC approval is limited to the installation in GV-SP Serial Number SN 5171 | 29.11.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|--------------------|----------------|----------------------|----------------|---|--|---|---|---------------|
| 10042618 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST04280AT-D | GV-SP | Installation of an Executive Completion | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI D-50 "Seat deformation into the overwing Emergency Exits and Emergency Egress." - (...) | FAA STC ST04280AT-D issued 19th November 2012-EASA/JAR Compliance Report, ref. GC528043076, Revision -, dated 20th November 2012-Installation in accordance with Gulfstream Aerospace Corporation Index List GC522048019, Revision -, dated 15th November 2 (...) | 1. The installer must determine whether this design change is compatible with previously approved modifications. 2. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person writt (...) | 11.12.2012 |
| 10042689 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST04249AT-D | G350 AND G450 | Installation of a Cabin Interior and Airplane Completion. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: Special Condition No. 25-450-SC (Gulfstream Aerospace Corporation, Model GIV-X Airplane (...) | Installation in accordance with Gulfstream Aerospace Corporation Index List GC420004225, Revision "-", dated January 19, 2012. Operation in accordance with Airplane Flight Manual Supplement, GC41204M015, Revision "-", dated January 20, 2012, is required (...) | 1. The installer must determine whether this design change is compatible with previously approved modifications. 2. The instructions contained in the Guidance for Follow-On Installation<->-> GC422040227, Revision A, dated January 19, 2012 or later F (...) | 13.12.2012 |
| 10042978 | GULFSTREAM AEROSPACE CORPORATION | EASA, IM.A.S.01259 | | FAA STC ST02796AT-D | GV-SP | Gulfstream Aerospace Corporation Modification FAA STC ST 02796AT-D. Installation of wireless LAN system. | | Gulfstream DAS Report Number GC51303K007 dated 26/07/05, Flight Manual Supplement ref GC 51 303M002 dated 06/06/05 - 503config & GC51303M003 dated 06/06/05 for - 505 config, or later approved revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 09.08.2005 |
| 10042985 | GULFSTREAM AEROSPACE CORPORATION | EASA, IM.A.S.01876 | | FAA STC ST02050CH-D | GIV-X | Installation of Class B Baggage Compartment. | | In accordance with Gulfstream Aerospace Corporation Master Drawing List GC425046085, Revision B, dated June 10, 2005, or later approved revisions, Gulfstream Flight Manual Supplement ref GC42 504M000 Rev - dated 10 June, 2005, or later approved revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 17.03.2006 |
| 10043072 | GULFSTREAM AEROSPACE CORPORATION | EASA, IM.A.S.01349 | | FAA STC ST02105LA-D | GV-SP (G550) | Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5221. | | - Install in accordance with Gulfstream Aerospace Report, GC52000K633, Rev. N/C and Index List GC520008643, Rev. G, dated 20 April 2009, or later approved revisions. - Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC52000M96, r (...) | Aircraft Serial Number 5221 only | 07.05.2009 |
| 10043073 | GULFSTREAM AEROSPACE CORPORATION | EASA, IM.A.S.01318 | | FAA STC ST03666AT-D | Data not found | Installation of an Airplane Interior and Completion for Aircraft Serial 5225. | | - Install in accordance with Gulfstream Aerospace Corporation Master Index List, GC514915225 Rev C, dated May 05, 2009 or later FAA approved revision. - To be operated in accordance with FAA approved Airplane Flight Manual Supplement, ref. GC51468M004, (...) | Aircraft Serial Number 5225 only. | 20.05.2009 |
| 10043733 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST04254AT-D | G650 | Broadband Multi-link (BBML) System STC incorporates the installation of a Broadband Multi-link (BBML) System in Gulfstream model GVI. The BBML system will provide an onboard broadband internet data connection using a satellite mobile broadband router, an antenna controller and a tail-mounted antenna. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA Statement of Compliance 100G-13-02, dated 14th of February 2013; Index List GDL100000004, Revision -, dated 16th of January 2013; Airplane Flight Manual GC61303M001, Revision -, dated 16th of January 2013; Instructions for Continued Airworthiness GC6 (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 20.02.2013 |
| 10043755 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST04277AT-D | GVI (G650) | Honeywell LSZ-860 Lightning Sensor System | The Certification Basis for the original product remains applicable to this certificate/ approval. | FAA Statement of Compliance letter 100G-13-030, dated 19/02/2013. Airplane Flight Manual Supplement GC61225M001, Revision -, dated 01/11/2012. Instructions for Continued Airworthiness GC61225A001, Revision -, dated 27/09/2013 or later revisions of the above (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 22.02.2013 |
| 10043757 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST04253AT-D | GVI (G650) | Honeywell AIS-2000 Satellite TV System | The Certification Basis for the original product remains applicable to this certificate/ approval. | FAA Statement of Compliance letter 100G-13-023, dated 19/02/2013. Airplane Flight Manual Supplement GC61507M001, Revision -, dated 26/10/2012. Instructions for Continued Airworthiness GC61507A001, Revision -, dated 26/10/2012 or later revisions of the above (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 22.02.2013 |
| 10044206 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST04261AT-D | GV-SP | Installation of Airplane Completion | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Finding(s); CRI D-50 Cabin Egress and Seat Encroachment. The requirements for environmental protection and the associated certified noi (...) | FAA STC ST04261AT-D, issued 27th June 2012; Gulfstream Aerospace Corporation Index List GC520005365, Revision -, dated 26th June 2012; GV-SP Airplane Flight Manual Supplement ref. doc. GC51204M006, Revision A, dated 15th August 2012- Instruction for Cont (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 26.03.2013 |
| 10044895 | GULFSTREAM AEROSPACE CORPORATION | | | FAA STC ST01926LA-D | GV-SP | Installation of 16-Passenger Cabin Interior with one additional crewseat. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI D-50 Cabin Egress and Seat encroachment. The requirements for environmental protection and the associated certified noi (...) | FAA STC ST01926LA-D, dated 27th February 2007; Gulfstream Aerospace Corporation Index List GC520008555, Revision G, dated 27th February 2007; GV-SP Airplane Flight Manual Supplement GC52000M039, original release, dated 27th February 2007; Instr (...) | This approval is limited to only the installation made in Gulfstream Model GV-SP Serial No. 5129. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and any other previously installed (...) | 15.05.2013 |
| 10045299 | GULFSTREAM AEROSPACE CORPORATION | | REV. 1 | FAA STC ST04252AT-D | G650 | G-VI Airplane Interior Installation - Description: Cabinetry, Seating, Associated Electrical and Mechanical Systems and Exterior Markings | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI D-01 "Shower Installation" The requirements for environmental protection and the associated certified noise and/or emis (...) | FAA Approval ST04252AT-D, Date of Issuance 28th of October 2012; FAA Statement of Compliance 100G-13-212, dated 17th of September 2013; Airplane Flight Manual Supplement - AFMS - GC51468M001, Revision B, dated 2nd of May 2013; Instructions for (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 18.09.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|-------------------|----------------|----------------------|------------------------|---|---------------------|---|--|---------------|
| 10015351 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.00567 | REV. 1 | FAA STC ST10016SC-D | GULFSTREAM 200 | Installation of BE Single (light weight) Seats Arrangement. | | -FAA STC ST10016SC-D-Index List GA222088001 Rev. D, dated 27/03/2003, or later FAA approved revision-FAA approved flight manual supplement GA22208M001, Rev. None, dated 27/03/2003 or later FAA approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 04.12.2008 |
| 10015593 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.01506 | | FAA STC ST399CH | BOMBARDIER CL-600-2B16 | Installation of a Honeywell FMZ-2000 Flight Management System | | -Gulfstream Aerospace Drawing List No 601-507-5077-2 Revision 1 dated 11 February 2002-Gulfstream AFM Supplement No CL6-AFMS-0015 Revision G dated 11 February 2002 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 12.09.2005 |
| 10015595 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.01508 | | FAA STC SA589GL | BOMBARDIER CL-600-2B16 | Installation of Passenger Oxygen System and Modification of Crew Oxygen System | | -K-C Aviation Inc Engineering Drawing List 600-017 Revision M dated 17 october 1989- AFM Supplement AFMS-26 for option 6 dated 27 August 1987 | 1. Canadian Part Numbers: 600-34012-73, located at B.L.6.06, and 600-34012-71, located at B.L.13, 73, must be replaced if any cracks are detected during the inspections required at intervals of 1000 take-off/landing cycles with this modification instal (...) | 12.09.2005 |
| 10015596 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.01509 | | FAA STC SA3716SW | BOMBARDIER CL-600-2B16 | Installation of a baggage compartment smoke detector system | | -K-C Aviation Inc Drawing List No T10-049 Revision C dated 3 September 1987 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 12.09.2005 |
| 10015597 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.01510 | | FAA STC SA4372SW | BOMBARDIER CL-600-2B16 | Installation of an emergency lighting system | | -K-C Aviation Inc Drawing List No T10-010 Revision D | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 12.09.2005 |
| 10015598 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.01511 | | FAA STC SA8353SW | BOMBARDIER CL-600-2B16 | Installation of a Loral F-1000 Flight Data Recorder | | -K-C Aviation Inc Drawing List No T10-075 Revision A dated 17 June 1992-K-C Aviation Inc AFMS-244 Initial Release dated 30 June 1992 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 12.09.2005 |
| 10015965 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02114 | | FAA STC ST09639SC | GULFSTREAM 200 | Installation of an Allied Signal DFDR i.a.w. MDL30-1100007 | | AFMS 30-1120011 | | 16.10.2006 |
| 10015966 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02115 | | FAA STC ST09945SC | GULFSTREAM 200 | Installation of Lead Acid Batteries i.a.w. MDL30-1100014 | | AFMS 30-1120013 | | 16.10.2006 |
| 10015994 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02148 | | FAA STC ST03270AT-D | GULFSTREAM 200 | Installation of a protective shroud around the 60Hz inverter | | Index List: CEB2 4938000, Rev. B or later-Instructions for Continued Airworthiness: CEB2 493A000 | | 21.11.2006 |
| 10016138 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02334 | REV. 1 | FAA STC ST02873AT-D | GULFSTREAM 200 | Installation of a Runway Awareness and Advisory System i.a.w. Index List CEB1 3498000. Note: This Revision corrects a typo in the Original Product Type Certificate Number. | | AFMS CEB1 349M001, AFMS CEB1 349M000 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 04.03.2009 |
| 10016182 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02411 | | FAA STC ST10021SC | GULFSTREAM 200 | Installation of an executive cabin interior on Gulfstream 200 S/N 065 including miscellaneous mechanical and electrical modifications to S/N 065. | | Index list GA220008065, rev. A, dated 17/05/2003-FAA approved Flight Manual Supplement GA 21468M065, Rev.-, dated 17/05/2003-Misc Mech Mod List GA244680065-Misc Elec Mod List GA214680065 or later revisions of the above listed documents approved by E (...) | This approval is limited to Gulfstream G200 S/N 065 only. | 15.08.2007 |
| 10016229 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02485 | | FAA STC ST10314SC-D | GULFSTREAM G500 | Installation of a Satcom Antenna. | | IFCA:GA34103A000 | | 09.10.2007 |
| 10016263 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02543 | | FAA STC ST10316SC-D | GULFSTREAM G150 | Installation of an Executive cabin interior in accordance with Gulfstream Index List GA322048000, Rev "-" dated 14 August 2006 or later FAA approved revision. | | FAA approved AFM Supplement GA31204M000, Rev. J, dated 12 November 2007 or later FAA approved revision. Instructions for Cont. Airworthiness. | Installation of this STC requires incorporation of 1) STC ST10314SC-D Installatio of ICG/Iridium Stacom Antenna, 2) When optional XM radiosystem is installed, XM Antenna installation is required per STC ST10313SC-D (both have to be EASA validated).Th (...) | 10.12.2007 |
| 10016264 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02544 | | FAA STC ST10317SC-D | GULFSTREAM G150 | Installation of a Collins IFIS 5000 EFB System. | | AFMS: GA31120M000 or GA31120M001IFCA: GA31120A000 | | 10.10.2007 |
| 10016265 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02545 | | FAA STC ST10321SC-D | GULFSTREAM G150 | Installation of a WX-1000 Stormscope | | AFMS: GA31225M000IFCA: GA31225A000 | | 09.10.2007 |
| 10016266 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02547 | | FAA STC ST10327SC-D | GULFSTREAM G150 | Installation of a Collins DBU 5000 System. | | AFMS: GA31320M000IFCA: GA31320A000 | | 09.10.2007 |
| 10016267 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02548 | | FAA STC ST10324SC-D | GULFSTREAM G150 | Installation of a second ADF. | | AFMS: GA31214M000IFCA: GA31214A000 | | 05.12.2007 |
| 10016268 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02549 | | FAA STC ST10326SC-D | GULFSTREAM G150 | Installation of a Honeywell FDR. | | AFMS: GA31420M000IFCA: GA31420A000 | | 09.10.2007 |
| 10016308 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02610 | | FAA STC ST10312SC-D | | Installation of a VHF COM Antenna. | | IFCA: GA34300A000. | | 22.11.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|-------------------|----------------|----------------------|-----------------|--|---------------------|--|---|---------------|
| 10016309 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02611 | | FAA STC ST10315SC-D | GULFSTREAM G150 | Installation of a 3rd VHF COM with Data link. | | IFCA: GA31300A000AFMS: GA31300M000 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 22.11.2007 |
| 10016334 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02643 | | FAA STC ST09823SC-D | GULFSTREAM 200 | Installation of Dual UNS 1C + upgraded FMS i.a.w. Index List CEB11188001 | | 1. AFMS CEB1118M0012. IFCA CEB1118A001 | 1. For A/C S/N004 through 051 only.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect (...) | 18.12.2007 |
| 10016336 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02645 | | FAA STC ST10329SC-D | GULFSTREAM 200 | Installation of a forward RH Galley with a convertible high Temp Oven/Microwave Oven Configuration. | | Installation of FWD right hand Galley with a convertible high Temp Oven/Microwave oven configuration in accordance with Index list GA223288002,Instruction for continued Airworthiness GA22328A001, dated 19.07.2007. | A prerequisite to the convertible Oven configuration is the installation of the Galley Multiple STC ST10304SC-D; installation limited to Gulfstream G200, S/N 163 only. | 17.04.2008 |
| 10016346 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02663 | | FAA STC ST10313SC-D | | Installation of an XM Antenna. | | ICA GA34607A00 | | 05.02.2008 |
| 10016364 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02686 | | FAA STC ST10319SC-D | | Installation of Laserref V IRS. | | ICA Ga31313A000AFMS GA31313M000 | | 25.03.2008 |
| 10016365 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02687 | | FAA STC ST10320SC-D | | Installation of 115 cubic foot oxygen cylinder in accordance with Index list GA327088000, Rev B, dated 15.09.2006 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2007/04/CF (or subsequent revisions of this decision). | | Instruction for Continued Airworthiness for Installation of A 115 cubic Foot Long range Oxygen Cylinder.Weight and Balance Statement G150-Instl of Long Range Oxygen Tank. | | 07.04.2008 |
| 10016406 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02735 | | FAA STC ST10334SC-D | | Installation of a Lavatory Smoke Detection system. | | ICA GA331455A000AFMS GA31455M000 | | 18.03.2008 |
| 10016486 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02842 | | FAA STC ST10296SC-D | GULFSTREAM 200 | Installation of a Cabin interior | | Index List GA220008102; rev A dated 1/12/2005 ;Flight manual supplement GA21468M102, Rev Noneor later revisions of the above documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | G-200, S/N 102. The compatibility of this design change with previously approved modifications must be determined by the installer . | 08.07.2008 |
| 10016487 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02843 | | FAA STC ST10295SC-D | GULFSTREAM 200 | Installation of a 3rd FMS 6100 Flight Management System i.a.w. Index List GA213188000, Rev C or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | | AFMS GA21318M000 Rev - or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 28.05.2008 |
| 10016488 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02844 | | FAA STC ST10294SC-D | GULFSTREAM 200 | Installation of a Pedestal Extension in a Gulfstream G200. | | Index List GA220048000, Rev A or later revision of the above document approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.The i (...) | 08.07.2008 |
| 10016556 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02940 | | FAA STC ST10338SC-D | GULFSTREAM G150 | Installation of an Emergency Power/Annunciator System i.a.w. Gulfstream Index List GA313318000, Rev. B or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | | ICA GA34331A000AFMS GA31331M000 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 29.07.2008 |
| 10016557 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02941 | | FAA STC ST10303SC-D | GULFSTREAM 200 | Installation of an XM Satellite Radio System i.a.w. Index list GA216078000, Rev. A. or later approved revisions of the aforementioned document approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | | ICA GA 24607A000 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 29.07.2008 |
| 10016584 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.02989 | | FAA STC ST10035SC-D | GULFSTREAM 200 | Installation of Executive Cabin Interior i.a.w. Gulfstream Index List GA220008063, Rev. A or later revisions of this document approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | | ICA GA22801A063AFMS GA21468M063 | Limited to S/N 063 | 20.08.2008 |
| 10016632 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.03069 | | FAA STC ST10291SC-D | GULFSTREAM 200 | Installation of Satcom High Speed Data Antenna i.a.w. IndexList GA243038000 Rev. D or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | | IFCA: GA24303A000 | 1 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product1 (...) | 21.01.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|-------------------|----------------|------------------------|--|--|---------------------|---|---|---------------|
| 10016633 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.03070 | | FAA STC ST10290SC-D | GULFSTREAM 200 | Relocation of AFIS/VHF Comm No. 1 Antenna i.a.w. Index List GA243068000 Rev. C or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | | IFCA: GA24306A000 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The In (...) | 22.01.2009 |
| 10016634 | GULFSTREAM AEROSPACE CORPORATION. | EASA.IM.A.S.03071 | | FAA STC ST10292SC-D | GULFSTREAM 200 | Installation of Thrane & Thrane Aero HSD+ Satcom System i.a.w. Index List GA213038000 Rev. D or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | | IFCA: GA21303A000AFMS; GA21303M000 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 22.01.2009 |
| 10014042 | GULFSTREAM AEROSPACE LIMITED | EASA.A.S.01120 | | | GULFSTREAM IV | Installation of Artex C406N Emergency Locator Transmitter | | Definition Document GAeL/0138/001 Issue 3 dated 25 January 2005 or later approved revision | | 15.02.2005 |
| 10014545 | GULFSTREAM AEROSPACE LIMITED | EASA.A.S.02632 | | | GULFSTREAM GV-SP | Certification of FAA Field Approved Modifications to enable the issue of a Certificate of Airworthiness on the Swiss register. Aircraft Serial No 5040 | | In accordance with Gulfstream Aerospace Limited Change Leaflet for Change No GAeL 06/02 Issue 2 | Aircraft Serial No 5040 Only | 24.08.2006 |
| 10014828 | GULFSTREAM AEROSPACE LIMITED | EASA.A.S.03041 | | | GULFSTREAM GIV-X | EASA Certification of a number of FAA Major Alterations installed at original delivery. | | Installation in accordance with GaeL modification No. GaeL 07/01, latest approved revision. Aircraft to be maintained in accordance with Instructions for continued Airworthiness ref - CE41801A002 Rev. - 17/9/07, or later approved revisions. To be opera (...) | Aircraft GIV-X Serial No. 4004 only. | 09.10.2007 |
| 10015024 | GULFSTREAM AEROSPACE LIMITED | EASA.A.S.03330 | | | GULFSTREAM GV-SP | Repositioning of Cabin Seat to meet the requirements of JAR25.562(c)(8). | | In accordance with Gulfstream Aerospace Limited Change Leaflet for Change No GAeL 07/06 Issue 1, dated 24 October 2007, or later approved revisions. | Aircraft Serial No 5032 Only. | 29.10.2007 |
| 10015276 | GULFSTREAM AEROSPACE LIMITED | EASA.A.S.03787 | | | GULFSTREAM GV-SP | Re-placard aft seats #Not for use during Taxi Takeoff & Landing#. | | Install in accordance with Gulfstream Aerospace Limited Change Ref GAeL 09/01. | 1. Aircraft Serials:- GV Serials 572 and 582;- GVSP Serial 5078.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification (...) | 18.02.2009 |
| 10015413 | GULFSTREAM AEROSPACE SERVICES | EASA.IM.A.S.01132 | | FAA STC ST2137CH-T | GULFSTREAM GV-SP | Custom 14 Passenger Interior & Aircraft Completion. | | Gulfstream MDL GC522046016 at latest FAA Approved Revision. Flight Manual Supplement GC52204-M005 dated July 14 2005 or later approved revision. | Limited to Gulfstream GV-SP Registration G-JC6B Serial Number 5060. | 21.02.2007 |
| 10015533 | GULFSTREAM AEROSPACE SERVICES | EASA.IM.A.S.01413 | | FAA STC ST00640CH | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 | Installation of the Honeywell Enhanced Ground Proximity Warning System | | -Gulfstream Aerospace Corporation Drawing List 601-5161-1, Rev N dated July 22, 2002 or later approved revision-FAA AFMS Supplement CL600-AFMS-00023 Rev E dated June 13, 2002 for MK V EGPWS without display | | 26.07.2005 |
| 10015688 | GULFSTREAM AEROSPACE SERVICES | EASA.IM.A.S.01673 | | FAA STC ST02229CH-D | GULFSTREAM GIV-X | Installation of Corporate Aircraft Completion STC | | In accordance with Gulfstream Aerospace Corporation Master Drawing List GC422046078, Revision A, dated February 28, 2006 or later approved revision Gulfstream Flight Manual Supplement ref GC42204M021 Rev - dated 28 February 2006, or later approved revision | 1. Aircraft Serial Number 4027 only 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 13.03.2006 |
| 10015848 | GULFSTREAM AEROSPACE SERVICES | EASA.IM.A.S.01948 | | ST10308SC-D | GULFSTREAM 100 GULFSTREAM 1125 WESATRD ASTRA GULFSTREAM ASTRA/SPX | Installation of Dual Universal Avionics UNS-1C+ Flight Management Systems. FAA STC 010308SC-D | | AFMS CEA1118M005IFCA CEA1118A000 | This STC is only applicable to A/C with previous installed dual Universal Avionics UNS-1C FMS. | 13.07.2006 |
| 10015856 | GULFSTREAM AEROSPACE SERVICES | EASA.IM.A.S.01960 | | FAA STC NO ST02237CH-D | GULFSTREAM 100 GULFSTREAM ASTRA/SPX | Installation of a Corporate Aircraft Completion in accordance with Gulfstream Master Drawing List GA122046000, Rev B dated July 19, 2006 or later approved revision | | IFCA ga11468A000 | Limited to S/N 158 | 11.08.2006 |
| 10014830 | H4 AEROSPACE (UK) Ltd. | EASA.A.S.03045 | | | BOEING B737-800 | Installation of a Satcom Antenna for a GSM system on B737-800 | | H4A891 issue 3, or later EASA approved revision-H4ACP043 issue 3, or later EASA approved revision-Instruction for continued Airworthiness H4ADR259 Issue 1 or later EASA approved revision-Weight and Balance report H4AWB203 Issue 2, or later EASA approved revision. | As in Instructions for Continued Airworthiness H4ADR259 Issue 1 or later EASA approved revision. | 16.01.2008 |
| 10014843 | H4 AEROSPACE (UK) Ltd. | EASA.A.S.03060 | | | | Interior Changes for the B/E Aerospace First Class Seat Integration in the Forward Passenger Cabin. This modification removes the existing 16 off Weber Business class seats and installs 12 B/E Aerospace First Class Passenger Seats with Ottomans and S (...) | | Modification Summary No. H4A887, Issue 1, dated 22nd May 2007. | 1. Instructions for Continued Airworthiness, Ref. No. H4ADR238.2. This approval is limited to only the installation made in the Boeing Model B767-306, Serial Number 30393 aircraft. 3. Prior to installation of this modification the installer (...) | 19.06.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|------------------|----------------|----------------------|---------------------------|---|---|---|---|---------------|
| 10014867 | H4 AEROSPACE (UK) Ltd. | EASA.A.S.03091 | | | BOEING B737-800 | Installation of GSM Phone / SATCOM system on B737-800 aircraft. | | Modification Summary Sheet: H4A911 Issue 3 or later EASA approved revision. Airplane Flight Manual Supplement: H4AFMS019 Rev. IR or later EASA approved revision. Additional Instructions for Continued Airworthiness: H4ADR331 or later EASA approved revision. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 12.12.2008 |
| 10014876 | H4 AEROSPACE (UK) Ltd. | EASA.A.S.03103 | | | BOEING 767-300 | The cabin reconfiguration of B767-300 Aircraft with Brice B1000S seats for TUI ARKFLY. This modification removes the existing 260 economy class seats and installs 260 new Brice B1000s economy class seats in an identical passenger layout. | | Modification Summary No. BBJ905, Issue 1, dated 16th May 2007. | Continued Airworthiness Manual, Ref. No. ML3263, Rev. A. This approval is limited to only the installation made in the Boeing Models B767-300, S/N 24477, S/N 24847, S/N 24848 aircraft. | 14.06.2007 |
| 10014883 | H4 AEROSPACE (UK) Ltd. | EASA.A.S.03111 | | | BOEING 737-500 | Implementation of Interior Changes associated with the cabin reconfiguration from 124YC to 126YC. | | Mod Summary H4A897 Iss.2. Instructions for Continued Airworthiness Doc. H4ADR383 Iss. 1. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The inst (...) | 10.06.2008 |
| 10015300 | H4 AEROSPACE (UK) Ltd. | EASA.A.S.03859 | | | AIRBUS A320-232 | GUEST/STAFF SEATING AREA RECONFIGURATION & MISC CABIN ITEMS FOR A320-232 AIRCRAFT as defined in H4ACP057. Model: Airbus A320-232 MSN 2403_A6-DLM | | # Modification summary and declaration of compliance H4A1196# Flight Manual Supplement H4AFMS019# Instructions for Continued Airworthiness H4ADR448 | Approval valid only for MSN 2403. The use of the aircraft is limited to non-commercial operation only. For other Limitations refer to Flight Manual Supplement H4AFMS019. | 04.05.2009 |
| 10026811 | H4 AEROSPACE (UK) Ltd. | P-EASA.A.S.03841 | | | BOEING 767-300 | Installation of Lifeport Stretcher. | As per TCDS. | Modification Summary Sheet, MOD H4A1197, Issue 2, 06/08/2009. Instructions for Continued Airworthiness, H4ADR446, Issue 1, 05/08/2009. | None. | 12.08.2009 |
| 10027495 | H4 AEROSPACE (UK) Ltd. | | REV. 1 | | B737-800 | Introduction of Recaro Economy Passenger Seats on Boeing 737-800, in accordance with project H4A1239 (Revision of STC 10027495 to remove EWIS Limitation) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01/Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane Systems (...) | Modification Summary Sheet: H4A1239 Issue 1 signed 02 October 2009. Instructions for Continued Airworthiness: H4ADR493 Issue 2 or later revisions of the above listed documents approved by EASA | None | 09.03.2011 |
| 10029103 | H4 AEROSPACE (UK) Ltd. | | REV. 2 | | A319, A320, A321 | Installation of Runway Turnoff and Logo Lights. Minor Change to EASA STC 10029103, Rev 1 issue date 31.03.2010 granted to H4 Aerospace (UK) Ltd. Minor change approval is sought to reflect the up issue wording of the instructions of continued airworthiness (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 / SC Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS. The c (...) | Modification Summary Sheet and Declaration of Compliance doc. N° H4A1269 Issue 5 dated 10/09/10. - Instructions for Continued Airworthiness doc. N° H4ADR512 Issue 3 dated 10/09/2010 and N° H4ADR513 Issue 3 dated 10/09/2010. - Master Document List doc. (...) | Restricted to the aircraft listed in doc. N° H4ADR530 - Master Effectivity List for Goodrich RTO and Logo Light Installation. | 20.10.2010 |
| 10032786 | H4 AEROSPACE (UK) Ltd. | | | | 737-700 | Interior Reconfiguration from 18 to 22 Passenger Capacity on Boeing 737-700 Aircraft | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS25 Amdt. 4 requirements as defined in the Certification Plan H4ACP062 Issue 6 dated (...) | Airplane Flight Manual Supplement Ref. H4AFMS020 Issue IR. Certification Plan Ref. H4ACP062 Issue 6 dated 04th October 2010. Modification Summary Ref. H4A1303 Issue 4 dated 21 October 2010 or later revisions of the above listed documents approved by EASA | None | 03.12.2010 |
| 10032969 | H4 AEROSPACE (UK) Ltd. | | REV. 1 | | BD-700-1A10 | Activation of Nicemate WLAN System | EASA.IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product areunch (...) | H4 Aerospace Modification Summary No. MOD H4A1334 Issue 4 dated 05 May 2011. H4 AFMS No. H4AFMS023H4 ICA No. H4ADR528 or later revisions of the above listed documents approved by EASA | Restricted to ASN 9011 | 24.06.2011 |
| 10033551 | H4 AEROSPACE (UK) Ltd. | | REV. 1 | | 737-300, -400, -500, -800 | Introduction of Cockpit Door Surveillance and Cabin Witness System | EASA TCDS IM.A.120 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are (...) | H4 Aerospace Modification Summary No. MOD H4A1356 Issue 2 dated 24 January 2011. H4 AFMS No. H4AFMS024H4 ICA No. H4ADR585 or later revisions of the above listed documents approved by EASA | None | 01.02.2012 |
| 10034127 | H4 AEROSPACE (UK) Ltd. | EASA.A.S.03836 | REV. 1 | | B737-800 | Introduction of Interior with 56 Recaro Business class seats and associated systems (Revision of STC EASA.A.S.03836 to remove EWIS Limitation) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: TGM/25/10 and TGL17 The certificated noise and/ or emissions levels of the original prod (...) | Modification Summary Sheet: H4A1165 Issue 2. - Instructions for Continued Airworthiness: H4ADR441 Issue 2 or later revisions of the above listed documents approved by EASA | The STC is limited to MSN 33016 (YJ935) and MSN 33021 (YJ936). - Changes to MMEL or MEL are not covered under the STC approval and require an additional approval, if deemed necessary. | 09.03.2011 |
| 10034368 | H4 AEROSPACE (UK) Ltd. | | | | B737-400 | Installation of a two-class seating arrangement | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Modification Summary Sheet and Declaration of Compliance Mod H4A1381 Issue 1 dated 14/03/2011 or later revisions of the above listed documents approved by EASA | Not Applicable | 29.03.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|--------------|----------------|----------------------|-----------------------------------|--|---|--|--|---------------|
| 10034588 | H4 AEROSPACE (UK) Ltd. | | | | 747-200, -300 | Installation of a Cabin/Vu-123 Cockpit Door Monitoring System and a Flight/Vu Witness Cabin Surveillance System in accordance with H4 Aerospace Modification Summary document nr. H4A1333 Rev. 1, dated 18/05/2010 or later approved revision. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H1 Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS. | H4AFMS022 Flight Manual Supplement H4ADR548 Instructions for Continued Airworthiness or later revisions of the above listed documents approved by EASA. | None | 13.04.2011 |
| 10035914 | H4 AEROSPACE (UK) Ltd. | | | | 777-200 | Introduction of an Aerospace Cockpit Door and Cabin Surveillance on Boeing 777-200. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-1-ICA on EWIS. The certificated noise and/ or emissions levels of the original product (...) | - H4A1332 Issue 2 dtd. 17. June 2011 Modification Summary- H4AFMS021 Issue IR dtd. 2. June 2011 Flight Manual Supplement- H4ADR547 Issue 1 dtd. 8. December 2010 Instructions for Continued Airworthiness or later revisions of the above listed documents approved by EASA. | none | 27.07.2011 |
| 10037544 | H4 AEROSPACE (UK) Ltd. | | | | B757-300 | Installation of Geven Puma Seats as defined in H4ACP076 | JAA TCDS JAA/25/98-019 Issue 2 The Certification Basis for the original product remains applicable to this certificate/ approval: CRI H01 EWIS. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of (...) | H4 Aerospace Modification Summary Sheet and Declaration of Compliance No. MOD H4A1469 Issue 1 dated 18 November 2011 or later revisions of the above listed documents approved by EASA. | None | 05.12.2011 |
| 10038071 | H4 AEROSPACE (UK) Ltd. | | | | 767-31K ER | Installation of Geven 'Nova' & 'Puma' Series Passenger Seats 326Y configuration. | EASA CS25 Change 4 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H01 Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS. | Modification Summary Sheet and Declaration of Compliance H4A1389 Issue 1 dated 04/01/12 or later revisions of the above listed documents approved by EASA. | Limited to installation on B767-31K Serial Numbers 27205, 27206 and 28865. | 16.01.2012 |
| 10038630 | H4 AEROSPACE (UK) Ltd. | | REV. 1 | | B777-200 | Installation of new cabin interior as defined in H4ACP075 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 EWIS. The requirements for environmental protection and the associated certificated (...) | H4 Aerospace Modification Summary and Declaration of Compliance No. MOD H4A1439, Issue 1, dated 02 March 2012 or later revisions of the above listed documents approved by EASA. | None | 10.07.2012 |
| 10040254 | H4 AEROSPACE (UK) Ltd. | | | | B757-200 | Passenger cabin reconfiguration on Openskies B757 | The original Certification Basis was updated by CRI H-01 (Enhanced Airworthiness Programme for Aeroplane system Instruction for Continued Airworthiness on EWIS) and CRI D-03 (Cabin layout-distribution of passenger seats # Interpretative Material) | H4ACP079 - Certification Plan H4ADR689 - Emergency Evacuation Review H4ADR724 - Enhanced Zonal Analysis procedure H4AFMS027 - Flight Manual Supplement 1757ME25202233 - Certification Review Drawing H4A1491 - Modification Summary H4ACCL552 - Compliance Checklist | None | 20.06.2012 |
| 10040531 | H4 AEROSPACE (UK) Ltd. | | | | 767-200, -300 | Introduction of an AD Aerospace Cockpit Door and Cabin Surveillance System on Boeing B767-200/300. | EASA.IM.A.035 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | H4 Aerospace Modification Summary and Declaration of Compliance No. MOD H4A1360 Issue 3, dated 06 July 2012 or later revisions of the above listed documents approved by EASA. | none | 09.07.2012 |
| 10041036 | H4 AEROSPACE (UK) Ltd. | | | | B757-200 | Cabin Re-configuration and Airvold IFE System Installation | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this approval. | H4 Aerospace Modification Summary Sheet, H4A1503, Issue 1, 16/08/2012 AFM Supplement, H4AFMS026, Issue IR, 16/08/2012 or later revisions of the above listed documents approved by EASA. | None | 17.08.2012 |
| 10043300 | H4 AEROSPACE (UK) Ltd. | | REV. 1 | | A330-202 | Stretcher installation and Aircraft cabin reconfiguration. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25 Amendment 4 - Various. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of (...) | EASA STC 10043300 H4 Aerospace Modification Summary and Declaration of Compliance No. MOD H4A1567 Issue 4 dated 10 March 2013 H4 Airplane Flight Manual Supplement No. H4AFMS028 H4 Instructions for Continued Airworthiness No. H4ADR739 or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 10.04.2013 |
| 10044860 | H4 AEROSPACE (UK) Ltd. | | | | 757-200 | Installation of Astronics EmPower (SPS system) [Installation of Astronics EmPower (SPS) system for European Aviation C2C20755 seats into B757-200 aircraft] | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: In addition to the original certification basis specified by TCDS EASA.IM.A.205 the applicant has demonstrated compliance (...) | Modification Summary and Declaration of Compliance, H4 Aerospace document No. H4A1483, Issue 5, dated 10 May 2013 - Aircraft Maintenance Manual Supplement, H4 Aerospace document No. H4AMS071, Revision 4, dated 10 May 2013 or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 13.05.2013 |
| 10045164 | H4 AEROSPACE (UK) Ltd. | | | | 1- A330-343 2- A340-311 / -313 | Introduction of LED based logo lights on Airbus A330 and A340 aircraft. This STC replaces the original logo lights by LED logo lights. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amdt. 4 (see details requirements through Certification Plan ref :H4CP072 issue 3) The requirements for environmental (...) | Modification summary sheet, ref : H4A1391 Issue 2 Instruction for Continued Airworthiness, ref : H4ADR611 Issue 1 or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 04.06.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|---------------------|----------------|----------------------|--|---|---|--|---|---------------|
| 10015668 | HARTZELL PROPELLER INC. | EASA.IM.A.S.01644 | | FAA STC SA01850CH | DIAMOND DA-40 | Installation of Hartzell Two blade propeller HC-C2YR-1BF/F7495S with STC SA01850CH in the Diamond DA40 aeroplane | | Hartzell engineering report No. 3085 Installation Instruction DA40-NST-EASAF Flight Manual Supplement DA40-AFMS-EASA | Diamond Modifications as detailed in the installation instructions | 30.09.2005 |
| 10030156 | HARTZELL PROPELLER INC. | P-EASA.IM.A.S.03250 | | FAA STC SA02677CH-D | PA-46-350P, PA-46R-350T | EASA validation of FAA STC SA02677CH-D Installation of Hartzell Propellers HCl3YR-1N/N7605(K)+2, HCl3Y1R-1N/N7605(K)+2 | CS-23.B NSC | Field Installation Instructions:1. INST_080808_97FLDfor PA-46-350P, S/N 4622001 to 4622200, 4636001 to 46361312. INST_080808_98FLDfor PA-46-350P, S/N 4636132 and upfor PA-46R-350T, S/N 4692001 and upFactory Installation Instructions:1. INST_080808_F (...) | For further limitations see applicable associated technical documentation | 31.05.2010 |
| 10034394 | HARTZELL PROPELLER INC. | | | FAA STC SA10551SC | 58, 58A, G58 | Installation of Hartzell PHC-J3YF-2UF/FFC7391D(B, K) propellers | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the certificated noise and/or emissions levels. | FAA approved Drawing No. 20002-0000, Rev. A, dated 20-Apr-2006. FAA approved Airplane Flight Manual Supplement AFMS20002-1, Rev. IR, dated 06-Jul-2006, is required for Baron 58, 58A, and G58 airplanes, serial number TH-1389, TH-1396 and after (Instruction ...) | Only aircraft equipped with a Teledyne Continental IO-550-C engine are eligible for this STC. | 31.03.2011 |
| 10038909 | HARTZELL PROPELLER INC. | | | FAA STC SA02130SE | KING AIR B200GT AND B200CGT | Installation of BLR Hartzell Composite Propeller HC-E4N-3A/NC9208K. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Certification basis defined on FAA STC SA02130SE. This certificate/ approval involves a (...) | Installation in accordance with Master Drawing List (MDL) Document BLR-KA250CP-002, Revision B, dated May 3, 2011, and Installation Instructions Document BLR-KA250006 Revision C, dated May 3, 2011. Operation in accordance with Airplane Flight Manual (...) | Requires the installation of a BLR Aerospace Ultimate Performance Package (UPP) ref FAA STC SA02131SE and EASA STC 10038758. | 22.03.2012 |
| 10039256 | HARTZELL PROPELLER INC. | | | FAA STC SA00470AT | A188, A188A, A188B | Installation of either a Hartzell model PHC-C3YF-1RF/F8468A(-) 6R propeller and associated spinner assembly C-3535-1(P) in accordance with FAA STC SA00470AT | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Bonaire Aviation Company Installation Instructions Propeller Installation in Cessna 188 series, Doc. No. BOAC 1010, dated January 17, 1994. Bonaire Aviation Company drawing Numbers 3.6 and 7, all dated January 17, 1994. FAA Approved Supplemental Airplane FI (...) | None | 20.04.2012 |
| 10015394 | HARTZELL SERVICE CENTER | EASA.IM.A.S.01095 | | SA1105CH | PIPER PA-32-260 | Installation of Hartzell HC-C3YR-1RF/F7693F three bladed propeller on the Piper-PA32-260 per FAA STC SA1105CH | | FAA approved AFM supplement PA-32-99-2 dated July 7, 1999 or later FAA approved revision | 1. EASA does not consider this modification to increase the noise level and it was not considered an acoustical change from the HC-C2YK-1F/F8477-2 Propeller. 2. If the holder agrees to permit another person to use this certificate to alter the product (...) | 30.05.2005 |
| 10015485 | HARTZELL SERVICE CENTER | EASA.IM.A.S.01245 | | FAA STC SA02181AK | CESSNA 208, CESSNA 208B | Installation of a Hartzell HC-B3TN-3AF(Y)/T10890CN(K)-2 three bladed propeller and D-4897(P) assembly on the type Cessna 208, 208A and 208B in accordance with the application dated 22 April 2005. FAA STC: SA02181AK | | Data package containing the following documents, and additional supporting data: 1. Noise test on SE-LZY, date April 22th 2006. 2. Noise certificate applied by the Swedish CAA for the SE-LZY, date May 3th 2006. 3. FAR Part 36 Noise Analysis by AeroTwin, Inc. (...) | -- | 07.08.2006 |
| 10015907 | HARTZELL SERVICE CENTER | EASA.IM.A.S.02033 | | SA02092AK | CESSNA 208 CESSNA 208A CESSNA 208B | Installation of Hartzell HC-D4N-3R/D9511F(K) Four Bladed Propeller and D-630-3(P) Spinner | | Installation in accordance with Aero Twin Inc Master Descriptive Data List No 208-4BP-D dated April 14, 1997, or later approved revisions. Operation in accordance with FAA approved Flight Manual Supplement No. 208-4BP-AFMS, dated October 17, 1997, or (...) | This approval is limited to Model 208 landplanes only. | 13.04.2007 |
| 10015985 | HARTZELL SERVICE CENTER | EASA.IM.A.S.02138 | | FAA STC SA01024CH | PIPER PA-28R-201T PIPER PA-28RT-201T | Installation of a Hartzell PHC-C3YF-1RF/F7663-4T propeller and C-3570(P) spinner in accordance with Hartzell installation instructions HP2834 dated February 3, 1999 or later approved revision (validation of FAA STC SA01024CH) | | FAA approved Airplane Flight Manual Supplement HP283-AFMS dated February 11, 1999 or alte approved revision, is required for PA-28R-201T (Turbo Arrow III) FAA approved Airplane Flight Manual Supplement HP284-AFMS dated February 12, 1999 or later appr (...) | The aircraft complies with annex 16, volume I, third edition, amendment 7, chapter 6. The certified noise level is 75.1dB(A) at 2575 rpm. This STC is approved only for the product configuration as defined in the approved design data referred to in the (...) | 09.11.2006 |
| 10016516 | HARTZELL SERVICE CENTER | EASA.IM.A.S.02887 | | FAA STC SA2078CH | CESSNA R172K | Installation of a Hartzell Propeller Inc. 75# Diameter 3-Blade Propeller model (P)HC-C3YF-1RF/F7392 with A-2295-1(P) spinner model (P)HC-C3YF-1RF/F7392 with C-4582-(P) spinner on a Cessna R172K (Hawk XP) Landplane aircraft. | | STC No. SA02078CH, dated December 3, 2004. Hartzell Document HP-R172K-NST, instructions for installation of a Hartzell (P)HC-C3YF-1RF/F7392 propeller with C-4582-(P) spinner (P)HC-C3YF-1RF/F7392 with A-2295-1(P) spinner on Cessna R172K #Hawk XP# at (...) | This STC is compatible with STC SA1437CE that installs the TCW IO-360-K, or #KB, engine as modified by STC SE1436CE. The re-rated engine is designated IO-360-KC, or #KBC, which increases the power rating to 210hp@2800 rpm for five minutes at takeoff (...) | 05.08.2008 |
| 10015337 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.00364 | REV. 2 | FAA STC SA2824SW-D | BAE 125 SERIES 800A HAWKER 800, 800XP, 850XP HAWKER HS 125-700 | Installation of a Long Range Oxygen System | FAA CAR 4b (<-&-> FAR Part 25 | Master Drawing List 125-012 Revision T or later approved revision. Operation in accordance with Flight Manual Supplement 800-250 Rev R6 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.02.2010 |
| 10015472 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01224 | REV. 3 | FAA STC ST10385C-D | HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of a Rockwell Collins CMS-1 Cabin Management System. | | 1. In accordance with Master Drawing List 800-512 Revision AB, or later approved revision; 2. Flight Manual Supplement 800-318 Revision R6 or later approved revision; 3. Maintenance Manual Addenda 33-20-10, 33-20-15, 23-30-02. | 1. This modification requires installation of STC ST10369SC-D. Installation of an Emergency Light System; | 21.05.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|----------------------|--|---|------------------------------|---|--|---------------|
| 10015473 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01225 | REV. 3 | FAA STC ST10369SC-D | HAWKER 750 HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of an Emergency Light System. | | 1. In accordance with :Master Drawing List 800-513 Revision L or later approved revision;2. Flight Manual Supplement 800-317 Revision R5 or later approved revision;3. Maintenance Manual Addenda 33-50-05. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 21.05.2008 |
| 10015644 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01608 | | FAA STC ST01342WI-D | HAWKER 800 XP | Installation of Collins Integrated Flight Information System (IFIS 5000) and Pro Line 21 Radios | | Collins Master Data List AAC-0360-001 Rev - , dated July 8,2004, or later approved revision.Collins airplane Flight Manual Supplement ref. AAC-0366-001, Rev - , dated June 6, 2005, or later approved revision.Collins Maintenance Manual Supplement ref. (...) | 1. Prior to installation and approval of Raytheon Aircraft Company Kit 149-3419-0001 is prerequisite to this STC.2. Prior to installation and approval of Rautheon Aircraft Company Kit 149-3420-0001 is prerequisite to this STC.3. This STC is a (...) | 19.10.2005 |
| 10015861 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01974 | REV. 2 | FAA STC ST10066SC-D | HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of B.F. Goodrich WX-1000E Stormscope. | | 1. In accordance with Master Drawing List 800-500 Revision G or later approved revision;2. Airplane Flight Manual Supplement 800-307 Revision R5 or later approved revision;3. Maintenance Manual Addenda: 34-43-10. | | 20.05.2008 |
| 10015862 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01975 | REV. 1 | | HAWKER 800 XP HAWKER 850 XP | Installation of a Davtron Clock | | Master Drawing List 800-503 Revision G or later approved revision;Airplane Flight Manual Supplement 800-309 Revision R2 or later approved revision;Maintenance Manual Addenda: 31-20-45 | | 23.06.2008 |
| 10015863 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01976 | REV. 2 | FAA STC ST10108SC-D | HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of a Cockpit Speaker Volume Recalibration. | | In accordance with : Master Drawing List 800-505 Revision IR or later approved revision | | 20.05.2008 |
| 10015864 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01977 | REV. 2 | FAA STC ST10109SC-D | HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of a Cockpit Speaker Inhibit Switch. | | 1. In accordance with : Master Drawing List 800-504 Revision L or later approved revision;2. Flight Manual Supplement 800-310 Revision R6 dated September 13, 2007 or later approved revision;3. Maintenance manual addenda 23-50-20. | | 21.05.2008 |
| 10015865 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01978 | REV. 2 | FAA STC ST10128SC-D | HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of a Precise Flight Pulselight System Activated by (existing) TCAS. | | 1. In accordance with : Master Drawing List 800-507 Revision D or later approved revision;2. Flight Manual Supplement 800-313 Rev R5 or later approved revision;3. Maintenance Manual Addenda: 33-40-15. | | 21.05.2008 |
| 10015866 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01979 | | ST10189SC | HAWKER 800 XP HAWKER 850 XP | Installation of Digital Passenger Briefing System | | Master Drawing List 800-511 Revision F or later approved revision;Flight Manual Supplement 800-316 Revision R2 or later approved revision;Maintenance Manual Addenda 23-33-21 | -- | 18.09.2006 |
| 10015867 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01980 | REV. 2 | FAA STC ST10187SC-D | HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of a Satellite Communications System. | | 1. In accordance with : Master Drawing List 800-509 Revision J or later approved revision;2. Flight Manual Supplement 800-314 Revision R5 or later approved revision;3. Maintenance Manual Addenda 23-16-20 | | 21.05.2008 |
| 10015868 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01982 | | ST10190SC | HAWKER 800 XP HAWKER 850 XP | Installation of CMS 400 Checklist Management System | | Master Drawing List 800-510 Revision J or later approved revision;Flight Manual Supplement 800-315 Revision R2 or later approved revision;Maintenance Manual Addenda 23-15-20 | -- | 19.09.2006 |
| 10015869 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01984 | REV. 3 | FAA STC ST09262SC-D | HAWKER 750, 800XP, 850XP, HAWKER 900XP | Installation of a Third and Fourth 750 Liter Oxygen Bottle in Aft Equipment Bay. | FAA CAR 4b <-&-> FAR Part 25 | -Installation in accordance with Master Drawing List 800-152 Revision For later approved revision.-Operation in accordance with Airplane Flight Manual Supplement 800-264Revision R7 or later approved revision.-Maintain in accordance with Maintenance Manual Addendum: 35-06-01. | -101 and -103 installation: Model Hawker 750; Hawker 800XP;Hawker 850XP-105 installation: Model Hawker 900XP | 18.02.2010 |
| 10015870 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01985 | REV. 4 | FAA STC ST09651SC | HAWKER 750 | Installation of Cabin Interior and Avionics Systems | | In accordance with: Master Drawing List 800-181 Revision A or later approved revision.For Hawker 800XP, Hawker 850XP, Hawker 900XP: Airplane Flight Manual Supplement 800-289 Revision R8 or later approved revision.For Hawker 750: Airplane Flight Ma (...) | 1. Hawker 750 with 238M530760-0017 and -0019 installations only, are limited to a maximum of 10 passengers.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other p (...) | 15.12.2008 |
| 10015871 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01986 | REV. 2 | FAA STC SA8063SW | BAE 125 SERIES 800A HAWKER 800 HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of Cabin Headliner/Window Panel Brackets. | | In accordance with Master Drawing List 800-116 Revision D dated May 16, 2006 or later approved revision. | | 20.05.2008 |
| 10015872 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01987 | REV. 2 | FAA STC SA8062SW | BAE 125 SERIES 800A HAWKER 800 HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of a Therapeutic Oxygen System. | | 1. In accordance with Master Drawing List 800-125 Revision B or later approved revision;2. Flight Manual Supplement 125-037 Rev 6 or later approved revision. | | 20.05.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|----------------------|---|---|---|--|--|---------------|
| 10015873 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01988 | REV. 2 | FAA STC SA7972SW-D | HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of Airborne Flight Information System (AFIS). | | 1. In accordance with : Master Drawing List 800-092 Revision S or later approved revision;2. Flight Manual Supplement 800-242 Revision 6 or later approved revision;3. Maintenance Manual Addenda: 34-61-05 | | 20.05.2008 |
| 10015874 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01989 | REV. 2 | FAA STC SA7944SW | BAE 125 SERIES 800A HAWKER 800 HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of a Cabin Oxygen System. | | 1. In accordance with Master Drawing List 800-107 Revision D dated August 28, 2006 or later approved revision.2. Maintenance Manual Addenda: 35-06-00 & 35-20-05 | | 20.05.2008 |
| 10015875 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01990 | REV. 1 | FAA STC SA3810SW-D | BAE 125 SERIES 800A HAWKER 800 HAWKER 800 XP HAWKER 850 XP | Installation of Precise Pulselite Control Unit | | 1. In accordance with: Master Drawing List 800-048 Revision Y or later approved revision.2. Flight Manual Supplement 800-231 Rev R103. Maintenance Manual Addenda: 33-40-05, 33-40-10 | 1. 103 Installation is for Models Hawker 800XP, Hawker 850XP and Hawker 900XP.2. All installations except -103 is for Models BAE 125 Series 800A, Hawker 800, Hawker 800XP | 19.11.2007 |
| 10015876 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01992 | REV. 2 | FAA STC SA2710SW | BAE 125 SERIES 800A HAWKER 800 HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP HAWKER HS 125 | Installation of Observer's Jump Seat. | | 1. In accordance with Master Drawing List 125-008 Revision L or later approved revision;2. Maintenance Manual Addenda: 25-10-15 | | 20.05.2008 |
| 10015877 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01993 | REV. 2 | FAA STC ST01342W1-D | HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of Collins Integral Flight Information System (IFIS 5000) and Proline 21 Radios. | | 1. In accordance with Master Data List AAC-0404-001 Rev B, or later approved revision;2. AFMS AAC-0406-001 R5, or later approved revision;3. Maintenance Manual Supplement AAC-0407-001 Rev -, or later approved revision. | 1. Prior installation and approval of Kit 149-3419-0001 is a prerequisite;2. Prior installation and approval of Kit 149-3420-0001 is a prerequisite; | 20.05.2008 |
| 10015878 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01994 | REV. 2 | FAA STC ST09772SC-D | BAE 125 SERIES 800A HAWKER 800 HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of a Smoke Detector System. | | 1. In accordance with : Master Drawing List 800-192 Revision E or later approved revision;2. Airplane Flight Manual Supplement 800-297 Revision R5 or later approved revision;3. Maintenance Manual Addendum 26-11-00. | | 20.05.2008 |
| 10015879 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01995 | | ST0998SC | HAWKER 800 XP HAWKER 850 XP | Installation of the Magnastar C-2000 Airborne Telephone System | | Master Drawing List 800-501 Revision J or later approved revision;Airplane Flight Manual Supplement 800-305 Revision R2 or later approved revision;Maintenance Manual Addenda: 23-16-05, 23-16-10 | | 19.09.2006 |
| 10015880 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01996 | REV. 2 | FAA STC ST09986SC-D | HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of an Airshow System. | | 1. In accordance with Master Drawing List 800-502 Revision H or later approved revision.2. Airplane Flight Manual Supplement 800-306 Revision R6 or later approved revision.3. Maintenance Manual Addenda: 23-33-35, 23-33-40, 23-33-41, 23-33-45, 23-33-50, 23-33-60, 23-33-61. | | 20.05.2008 |
| 10015881 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.01997 | REV. 3 | FAA STC ST10001SC-D | 750, 800XP, 850XP, 900XP | Installation of a Honeywell Solid State Flight Data Recorder with interface to Collins Proline 21. | FAA CAR 4b & 14 CFR Part 25The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original p (...) | 1. Installation of the Honeywell Solid State Flight Data Recorder with interface to Collins Proline 21 in accordance with the Hawker Beechcraft STC Master Drawing List 800-199, Revision AJ.2. For -101, -105, and -107 Installations: Models: Hawker 750, (...) | For Hawker 800XP model, installation is limited to aircraft equipped with Collins Proline 21 Avionics | 19.09.2011 |
| 10016024 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.02192 | REV. 2 | FAA STC ST10432SC-D | HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of Mark V Enhance Ground Proximity Warning System (EGPWS) with and without Runway Awareness Advisory System. | | 1. In accordance with Master Drawing List 800-514 Revision D or later approved revision;2. Flight Manual Supplement 800-329 Revision R6 or later approved revision;3. Maintenance Manual Addenda 34-40-40. | 1. The Flight Manual Supplement is not needed for the -101 installation because the aircraft's basic AFM is still applicable.2. When fitted to Hawker 800XP and 850XP this STC is limited to the following aircraft 258541, 258556, 258567 through (...) | 21.05.2008 |
| 10016179 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.02408 | REV. 2 | FAA STC ST09956SC-D | BAE 125 SERIES 800A HAWKER 800 HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of a Divan Cargo Net Assembly. | | | 1. In accordance with Master Drawing List 800-197, Revision D, or later approved revision.2. Maintenance Manual Addenda 25-50-05, | 20.05.2008 |
| 10016598 | HAWKER BEECHCRAFT CORPORATION | EASA.IM.A.S.03011 | REV. 4 | FAA STC ST 10596SC-D | 800,750, 800XP, 850XP, 900XP | Installation of an Aircell Access Voice/Data Communications System. | As defined on FAA STC ST10596SC-D FAA CAR 4b & 14 CFR Part 25 per EASA TCDS IM.A.085 | In accordance with: STC Required Documents List (RDL) No. 230-2001 Rev.37 dated September 23, 2011 or later approved revision; Aircraft Flight Manual Supplement No. 230-3001 Rev. R9 dated September 23, 2011 or later approved revision, or 230-3002 Rev. (...) | This modification is only applicable to aircraft equipped with the Collins Proline 21 or Honeywell SPZ-8000 avionics packages.RDL 230-2001-103 and -107 Installation requires the aircraft be equipped with Collins Proline 21 radios with the IFIS STC ST01 (...) | 30.10.2012 |
| 10029917 | HAWKER BEECHCRAFT CORPORATION | | | FAA STC ST10599SC-D | 800XP, 850XP, 900XP | Installation of an Espresso Maker | FAA CAR 4b <-&-> 14 CFR Part 25 | Installation of an Espresso Maker in accordance with STC Required Documents List (RDL) No. 230-2003 Revision 5 dated January 14, 2010, or later approved revision. | This modification requires the installation of FAA STC ST10385SC-D.Installation of a Rockwell Collins CMS-1 Cabin Management System. | 03.05.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|--------------------------|----------------|----------------------|---|--|--|---|---|---------------|
| 10031606 | HAWKER BEECHCRAFT CORPORATION | P- EASA, IM.A.S.03021 | REV. 2 | FAA STC ST10922SC-D | MODEL 4000 | Installation of Cabin Interior Furnishings and Systems. | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this approval. | Master Documents List (MDL) No. 240-2001 Rev.69 Configuration 3: Configuration Drawing 240-044300 Rev.3 Airplane Flight Manual Supplement (AFMS) 240-590001-0007 Rev.3 Completions Manual 23-10-65- HBC10003P dated August 2010 and 44-30-20-HBC10009 dated J (...) | Only configuration 3, 6, 8 and 9 installations are EASA approved. Airworthiness Limitations inspections are contained in Chapter 4 of the Completions Manual. Airworthiness Limitations for configurations 6 and 8 per 240-590011-0003C approved August 2010; (...) | 16.10.2012 |
| 10032246 | HAWKER BEECHCRAFT CORPORATION | | | FAA STC ST10598SC-D | 4000 | Installation of Flight Camera System | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | STC Required Documents List (RDL) No. 240-2002-1, revision 11, October 13, 2009. Completions Manual Chapter 44-20-30-H4000, August 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 19.10.2010 |
| 10042768 | HAWKER BEECHCRAFT CORPORATION | | | FAA STC SA00782W1-D | EASA APPROVED MODEL LIST | Modification of Collins VHF-22A/VHF-22B VHF COMM Receiver Transmitter with CTL-22 Control Pro Line II System Installation to a Collins VHF-22C/VHF-22D VHF COMM Receiver Transmitter with CTL-22C Control 8.33KHz Pro Line II System Installation. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Collins Installation Instructions Drawing List no. AAC-0178-001 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | A VHF-22C unit can be installed to replace either an existing VHF-22A or VHF-22B unit. A VHF-22D unit can only be installed to replace an existing VHF-22B unit. | 18.12.2012 |
| 10015370 | HAWKER BEECHCRAFT CORPORATION (HBC) | EASA, IM.A.S.01009 | | FAA STC SA8356SW | HAWKER 1000 | Installation of Air Distribution System | | -FAA STC SA8356SW-Arkansas Aerospace Inc. Drawing List 1000-015-3 rev B dated June 21 1994 or later approved revision-AFM Supplement ref. 1000-0014 dated June 26 1992 or later approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.02.2005 |
| 10015373 | HAWKER BEECHCRAFT CORPORATION (HBC) | EASA, IM.A.S.01026 | | SA8348SW | HAWKER 1000 | Installation of Honeywell SRZ-8000 Digital Integrated Flight Control System with low speed awareness cue in accordance with Arkansas Aerospace Inc. Drawing List 1000-008 | | STC Drawing List 1000-008AFMs 1000-0009, 1000-0010, 1000-0015, 1000-0016, 1000-0038, 1000-0046 | This STC does not give PRNAV approval/RVSM approval/CAT II approval | 01.02.2007 |
| 10015909 | HAWKER BEECHCRAFT CORPORATION (HBC) | EASA, IM.A.S.02036 | | ST09250AC | HAWKER 1000 HAWKER 800 HAWKER 800 XP HAWKER 850 XP | Installation of a Logo Light on the lower surface of the Horizontal Stabiliser | | Installed in accordance with DevOre Top Drawing No 12500003, dated october 30, 1996 or later approved revision | | 21.09.2006 |
| 10015982 | HAWKER BEECHCRAFT CORPORATION (HBC) | EASA, IM.A.S.02134 | REV. 3 | FAA STC ST00459NY | HAWKER 750 HAWKER 800 HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of an Allied Signal (Garrett) GTCP 36-150W gasturbine Auxiliary Power Unit (APU). | | 1. Installation in accordance with - PATS, Inc Installation Master drawing List No RH1001, Rev E, dated December 12, 1997, or PATS Inc Master Drawing List No RH1001, Rev G, dated May 31, 2006, and PATS, Inc Master Drawing List RH5001, Rev G, d (...) | | 16.06.2008 |
| 10016127 | HAWKER BEECHCRAFT CORPORATION (HBC) | EASA, IM.A.S.02321 | | FAA STC SA01417W1 | RAYTHEON A36 | Replace pilot's altimeter with IS&S encoding altimeter. | | 1. RAS Master Data List Replace RAS2306009-40, Revision 6.0, dated December 7, 2006, or later approved revisions. 2. Airplane Flight Manual Supplement, RAS230600958, dated December 13, 2006, or later approved revisions | Compliance with the following requirements, in addition to the requirements shown on the Type Certificate Data Sheet 3A20, has been shown: 23.303, 23.305, 23.307, 23.321, 23.341, 23.365, 23.601, 23.609, 23.627 effective February 1, 1965 23.611 as amend (...) | 26.02.2007 |
| 10016230 | HAWKER BEECHCRAFT CORPORATION (HBC) | EASA, IM.A.S.02492 | | FAA STC SA0859W1 | BEECH 1900D | Installation of EROS MAGIC (Mask and Goggles Integrated Concept) Quick Donning Masks. | | Installation in accordance with Raytheon Aircraft Services Master Drawing List 148-100057, Revision A, dated January 27, 2000, or later approved revision. | | 17.07.2007 |
| 10016231 | HAWKER BEECHCRAFT CORPORATION (HBC) | EASA, IM.A.S.02494 | | FAA STC SA0876W1 | BEECH 1900D | Installation of an Allied Signal CAS 67A Traffic Alert and Collision Avoidance System (TCAS II) | | Raytheon Aircraft Services STC Master Drawing List No. 148-100040, Revision C, or later approved revision. Raytheon Aircraft Services Airplane Flight Manual Supplement, Document No. FMS002193UE363, dated April 27, 2000, or later approved revision. | | 17.07.2007 |
| 10016367 | HAWKER BEECHCRAFT CORPORATION (HBC) | EASA, IM.A.S.02689 | | FAA STC SA01798SE | BEECH 200 BEECH 200C BEECH 200CT BEECH 300 BEECH 300C | Reduced Vertical Separation Minimum Qualification Document Group Aircraft Classification Hawker Beechcraft Models B200/B200C/B200GT/B200CGT/B300/B300C (Super King Air) | | Aero-Mech AMI-AFMS-KAIR-0719 Airplane Flight Manual Supplement for Hawker Beechcraft Models B200/B200C/B200GT/B200CGT Aero-Mech AMI-AFMS-KAIR-0722 Airplane Flight Manual Supplement for Hawker Beechcraft Models B300/B300C | | 18.04.2008 |
| 10016577 | HAWKER BEECHCRAFT CORPORATION (HBC) | EASA, IM.A.S.02975 | | FAA STC ST02113LA | BEECH 400A | Rockwell Collins dual Attitude Heading Reference system (AHRs). | | None. | Cutter aviation Master Data List Document No. 02A-2406-201, Revision #B# dated December 5, 2007 or later approved revisions. | 12.09.2008 |
| 10030074 | HAWKER BEECHCRAFT CORPORATION (HBC) | | | FAA STC SA01566W1-D | KING AIR B200 SERIES | Installation of modified Rockwell Collins Flight Guidance Computers (FGC-3000 units) as a solution to eliminate nuisance pitch trim wheel movement. | FAR 23. The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Instructions for Replacement of FGC-3000 in Hawker Beechcraft Corporation King Air Models B200/B200C/B200GT/B200CGT, in accordance with Rockwell Collins Inc Master Drawing List ref P/NRCA-0241, Revision - , dated September 15, 2009. Instru (...) | Prior approved installation of a Rockwell Collins Inc FGC-3000 Flight Guidance Computer Module (CPN 822-1108-031) is a prerequisite to this STC. | 19.05.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------|----------------|----------------------|----------------------------|--|---|---|--|---------------|
| 10014008 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.01051 | REV. 1 | | HAWKER HS 125-700B | Increase Maximum Zero Fuel Weight | | Instructions for Continued Airworthiness ACS-R-007-711422.Weight Statement CSWE/L/800. | | 22.02.2005 |
| 10014102 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.01206 | | | BEECH 400A (HAWKER 400 XP) | Certification for Steep Approaches | | -Raytheon Aircraft Services Change SC0252A-Raytheon Aircraft Services AFM Supplement reference 128-59001-167 | | 26.07.2005 |
| 10014103 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.01207 | | | BAE 125 SERIES 800B | Raytheon Aircraft Services Modification SC0135R. Runway Awareness & Advisory System (RAAS) to the MK VII EGPWS Computer | | RAS Certification Plan ASC-C-008-AW1499 Issue 1 dated February 2005 or later approved revision.RAS AFM Supplement 16 Issue 1 to AFM ref HS.1.15 | | 21.06.2005 |
| 10014104 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.01208 | | | HAWKER HS 125-700B | Installation of a Collins AHS-3000A Attitude Heading Reference System. | | Certification Plan ASC-C-007-AV1490 Issue 1, dated February28, 2005 or later EASA approved revisions.Change SC0194U dated May 20, 2005 or later EASA approved Revisions. | | 13.06.2005 |
| 10014105 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.01209 | | | HAWKER HS 125-700B | Installation of Traffic Alert & Collision Avoidance System (TCAS-4000), Mode S Elementary Surveillance (Flight ID) and Emergency Locator Transmitter (Artes C406-2) | | Certification Plan ASC-C-007-AV1518 Issue 1 Dated February 18 2005 or later approved revision | | 23.03.2005 |
| 10014157 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.01283 | | | BAE 125 SERIES 800B | Special Interior Refurbishment | | - Certification Plan ASC-C-008-AW1476 Issue 1 Dated February 2005 or later approved revision.- Instructions for Continued Airworthiness ASC-008-AW1484 Issue 1 dated April 18, 2005 or later approved revision.- Change number. SC0246A. | | 28.04.2005 |
| 10014350 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.01615 | | | BAE 125 -800 SERIES | Steep Approach Capability | | Raytheon Aircraft Services Certification Plan ASC-C-008-AV1253.Raytheon Aircraft Services AFM Supplements Flight ManualRef. HS.1.16 Supplement 10 Issue 1 & HS1.15 Supplement 15 Issue 1 or laterapproved revisions. | | 25.10.2005 |
| 10014466 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.02527 | | | BAE 125 SERIES 800B | To introduce completion of the upgrade to the existing Terrain Awareness Warning System (TAWS) installation to include terrain display. | | Raytheon Aircraft Service Modification SC0133JRaytheon (AFMsupp ref H.S. 1.15 Supplement No. 18 | Applicable to Aircraft Serial 258208 only | 02.05.2006 |
| 10014803 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.03006 | REV. 1 | | HAWKER 850 XP | Introduction of Steep Approach Capability. | | Modification SC0281A AFMS Hawker Beechcraft Supplement 1 issue 1 or later approved issue. | | 03.07.2007 |
| 10015118 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.03467 | | | HAWKER 800 XP | To upgrade Dual Honeywell NZ-2000 FMS Computers from 5.2 to6.0 software and provide P-RNAV approval. | | Hawker Beechcraft Limited Change No. SC0314A # To introduceDual Honeywell NZ-2000 FMS Computers (with V6.0 software) in lieu of NZ-2000(with V5.2 software)Operation to be in accordance with Hawker Beechcraft Limited Flight Manual Supplement No.4 to (...) | None. | 06.10.2008 |
| 10015119 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.03468 | | | HAWKER 1000 | To introduce Dual Honeywell NZ-2000 FMS Computers in lieu of NZ-920 FMS Computers and add a second GPS GNSSU. The existing Data Loader is replaced with Model DL-950. | | Certification Plan ASC-C-009-AW3173. Instructions for continued airworthiness ASC-R-009-AV3152. AFM Supplement for Hawker 1000B (HS1.18) Supplement 3 issue 1.Modification ref SC0315A | 1. As per AFM Supplement for Hawker 1000B (HS1.18) Supplement 3 issue 1.2. Prior to installation of this modification the installer must determine that the interrelationshipbetween this modification and any other previously installed modifcat (...) | 25.09.2008 |
| 10015126 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.03476 | REV. 1 | | HAWKER 750 | Installation of Steep Approach Capability | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Aircraft with Honeywell EGPWS installed- Installed to Major Change SC0316A at issue 1 or later.- Operated to Airplane Flight Manual Supplement 1 for Flight Manual140-590039-0005.-Aircraft Crew Manual Supplement - Crew Manual - Series 750.-Maintained i (...) | None | 05.05.2011 |
| 10015127 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.03477 | REV. 1 | | HAWKER 900 XP | Installation of Steep Approach Capability | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Aircraft with Honeywell EGPWS installed- Installed to Major Change SC0317A at issue 1 or later.- Operated to Airplane Flight Manual Supplement 1 for Flight Manual140-590037-0005.-Aircraft Crew Manual Supplement - Crew Manual - Series 900XP.-Maintained (...) | None | 05.05.2011 |
| 10015133 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.03486 | | | HAWKER HS 125-700A | To upgrade Dual UNS-1K FMS to Dual UASC UNS-1Lw in preparation for P-RNAV compliance. | | ¿ HBL Major Change SC0318A¿ Certification Plan ASC-C-009-AW3264¿ HBL Supplement No.7 to Flight Manual H.S.1.9¿ ASC-Z-007-AV3270 Instructions for Continued Airworthiness. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 10.11.2008 |
| 10015322 | HAWKER BEECHCRAFT Ltd. | EASA.IM.A.S.00275 | | FAA STC 000983CH | BEECH 400A | Installation of a Honeywell Enhanced Ground Proximity Warning System (EGPWS) Mark VIII in accordance with Elliott Aviation Master Data List 2018-571DL, Revision 4, dated 6 November2003 or later EASA approved revisions.This STC is issued asresult of 1 (...) | | | 1. AFM Supplement Elliott Aviation Document No. FS2018-571 rev.0 or later EASA approved revisions.2. Instructions for Continued Airworthiness Document No. CA2018-571.3. This STC is approved only for the product configuration as defined in (...) | 20.10.2004 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------|-------------------|----------------|--------------------------|---|---|--|--|---|---------------|
| 10015332 | HAWKER BEECHCRAFT Ltd. | EASA.IM.A.S.00350 | | FAA STC SA00421W1 | BEECH 1900C, AIRLINER | Defines an all cargo interior using web nets. The all cargo interior is defined by the top drawing of Raytheon Aircraft Services No114-0100, no change, dated April 17, 1996, with an "FAA Approved" stamp of April 17, 1996 or later FAA-approved revision. | | | 1. The Airplane Flight Manual Supplement No RAS 114-590021 is required equipment when this STC is installed. 2. Compatibility of this design change with previously approved modifications must be determined by the installer. 3. This STC is iss (...) | 03.12.2004 |
| 10032984 | HAWKER BEECHCRAFT Ltd. | | | | B200, B300 | Installation of Thrane & Thrane Aviator 200/300 Inmarsat Satcom with Wi-Fi capability on Beechcraft King Air B200/B300 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installed to Major Change SC0353A at issue 1. Operation in accordance with Hawker Beechcraft Limited Airplane Flight Manual Supplements: - King Air B200: 101-590168 Supplement 1, Issue 1. King Air B300: 130-590031 Supplement 1, Issue 1. Aircraft to be maint (...) | This modification is only applicable to aircraft equipped with the Collins Proline 21 avionics package. Refer to aircraft defined in Hawker Beechcraft major change SC0353A. | 13.12.2010 |
| 10037296 | HAWKER BEECHCRAFT Ltd. | | REV. 1 | | 750, 800XP, 850XP, 900XP | To introduce interior and electrical provisions for Medevac stretcher/erole change capability. This Change instructs the interior and electrical changes to the aircraft. Stretcher and medical equipment installation will be certified by the appropriate vendor. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installed in accordance with Major Change SC0399A at issue 8. Maintained in accordance with Instructions for Continued Airworthiness document ref ASC-R-023-AW4014 at issue 3. Aircraft Flight Manual Supplement 140-590037-0005 Supplement 5 issue 1 or later (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 14.03.2013 |
| 10038135 | HAWKER BEECHCRAFT Ltd. | | | | HAWKER 750, 800XP, 850XP HAWKER 900XP | To introduce Dual L3 Electronic Flight Bags (EFB) units in the cockpit with power supplies. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installed to Major Change SC0362A at issue 4 or later. - Hawker Beechcraft Limited Aircraft Flight Manual Supplement 3 to AFM 140-590039-0005 (Hawker 750), or Supplement 3 to AFM 140-590032-0005 (Hawker 800XP), or Supplement 2 to AFM 140-590035-0005 (Ha (...) | All EFB functionality and software approval will be the responsibility of the aircraft operator. Unless approved otherwise by all Required Manuals (Hard Copy) must still be carried in the aircraft. | 23.01.2012 |
| 10038252 | HAWKER BEECHCRAFT Ltd. | EASA.A.S.03559 | | | HAWKER 750 HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | To introduce a new forward cabin door with associated placards. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. FAR 25.813(f) Amendment 25-17. The requirements for environmental protection and the asso (...) | Installed in accordance with Major Change SC0319A at issue 2 dated 23Jan 2012. Hawker Beechcraft Limited Aircraft Flight Manual Supplement 4 to AFM 140-590039-0005 (Hawker 750), or Supplement 5 to AFM 140-590032-0005 (Hawker 800XP), or Supplement 2 to AF (...) | None. | 01.02.2012 |
| 10038800 | HAWKER BEECHCRAFT Ltd. | | | | KING AIR B300CER | To introduce increased maximum landing weight to 16,500lbs for the King Air B300CER | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installed in accordance with Hawker Beechcraft Limited Major Change ref SC0409A at issue 1. Operation in accordance with Hawker Beechcraft Limited Aircraft Flight Manual Supplement 2, Issue 1 to AFM ref 130-590031-245. Maintained in accordance with Instr (...) | None. | 14.03.2012 |
| 10040927 | HAWKER BEECHCRAFT Ltd. | | | | C90GTI | To introduce L3 Electronic Standby Instrument System (ESIS) in lieu of existing Meggit Mk II Secondary Flight Display (SFD) System. The existing Meggit Mk II SFD (part number 39221-01-01) is removed and replaced with L3 GH-3900.2 ESIS (Part number 9200-34500-0401). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Hawker Beechcraft Ltd., UK <(>-<)> Change document no SC0413A at issue 3 <(>-<)> - Hawker Beechcraft Ltd., UK, Instructions for Continued Airworthiness document no ASC-R-018-AW4121 at issue 1 <(>-<)> - Hawker Beechcraft Ltd., UK, Aircraft Flight Manual Supp (...) | None | 03.08.2012 |
| 10041476 | HAWKER BEECHCRAFT Ltd. | | REV. 1 | | B200 B200GT | Introduce Rockwell Collins TCAS 4000, Traffic Collision Avoidance System (TCAS) II System. | FAR-23, per TCDS. The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are (...) | Change No. SC0427A, Issue 1, July 2012; ICA ASC-R-018-AW4178, Issue 1, August 2012; or later revisions of the above listed documents approved by EASA. | STC does not provide compliance with Commission Regulation (EC)1332/2011 as regards to ACAS II equipment. | 05.10.2012 |
| 10016247 | HAWKER BEECHCRAFT SERVICES | EASA.IM.A.S.02524 | | FAA STC ST02619AT | BEECH 400 BEECH 400A | Installation of a Battery Low Voltage Warning / Shutoff System on Raytheon Aircraft 400 and 400A. | | Master Drawing List MDL RAF01106A, dated 10 June 2002 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | | 28.08.2007 |
| 10016248 | HAWKER BEECHCRAFT SERVICES | EASA.IM.A.S.02525 | | FAA STC SA03292AT | BEECH 390 | Modification to install an Auxiliary Ground Heating System. | | RAS Master Data List: RAS110500103, Rev B, dated 21 June 2006 or later FAA approved Revision. FAA approved RAS AFM-S Doc. RAS110500111, dated 29 August 2005, or later approved Revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The i (...) | 18.10.2007 |
| 10016250 | HAWKER BEECHCRAFT SERVICES | EASA.IM.A.S.02527 | | FAA STC NUMBER ST01441W1 | HAWKER 800 HAWKER 800 XP | Hawker Beechcraft Corporation Installation of upper and lower LED beacons, wingtip LED navigation lights and wingtip strobe lights. | | Installation as defined in Hawker Beechcraft Master Data List RAS050600152 rev 3.0, dated May 11, 2007, or later approved revisions. Instructions for Continued Airworthiness as defined in: - Hawker Beechcraft (ICA) document ref. RAS050600161, latest approved revision. | | 10.10.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|---------------------|----------------|----------------------|---|--|--|--|---|---------------|
| 10016269 | HAWKER BEECHCRAFT SERVICES | EASA.IM.A.S.02552 | | FAA STC ST03405AT | BEECH 400 BEECH 400A | Replacement of a Standby Altimeter installation in the Raytheon (Beech) models 400, 400A aircraft. This STC is a duplicate of STC ST03379AT, EASA.IM.A.S.02353 | | Installation in accordance with Master Data List (MDL), RAS1106DL004, Revision 1, dated January 4, 2007 or later approved revision. | | 04.09.2007 |
| 10016281 | HAWKER BEECHCRAFT SERVICES | EASA.IM.A.S.02574 | REV. 1 | FAA STC ST01369AT | BEECH 300 | Installation of a 70% Ground Idle Stop Dropout System. Installs cams and switches on the existing aircraft's power levers system to allow the propeller system to transition from Ground Idle to Flight Idle when both Power levers are advances above 70 %. (...) | | Installation in accordance with Raytheon Aircraft Services Master Drawing List Number MDL-TAF96106, Original Revision, dated February 12, 1997, or later FAA approved revision. Operation in accordance with Raytheon Aircraft Services Aircraft Flight Man (...) | | 19.10.2007 |
| 10016352 | HAWKER BEECHCRAFT SERVICES | EASA.IM.A.S.02672 | | FAA STC SA01414W1 | HAWKER BEECHCRAFT G58 (BARON) | Modification to install a Honeywell KR 87 Automatic Direction Finder (ADF) System and the Honeywell KA-44B Antenna interfaced with the previously installed Garmin 1000 Avionics System. | | Hawker Beechcraft Services:# Master Drawing List No. RAS050600084 Revision 1.0 dated 11 Oct. 2006 or later FAA approved revisions# Airplane Flight Manual Supplement RAS050600079 dated 6 Oct. 2006 or later FAA approved revision. | None | 27.05.2009 |
| 10016353 | HAWKER BEECHCRAFT SERVICES | EASA.IM.A.S.02673 | | FAA STC SA01455W1 | HAWKER BEECHCRAFT G36 (BONANZA) | Modification to install a Honeywell KR 87 Automatic Direction Finder (ADF) System and the Honeywell KA-44B Antenna interfaced with the previously installed Garmin 1000 Avionics System. | | Hawker Beechcraft Services:# Master Drawing List No. 002-3450-L01-00 Revision 1.0 dated 26 Sept. 2007 or later FAA approved revisions# Airplane Flight Manual Supplement 002-3450-A01-00 dated 15 Oct. 2007 or later FAA approved revision. | None | 27.05.2009 |
| 10016550 | HAWKER BEECHCRAFT SERVICES | EASA.IM.A.S.02930 | | FAA STC ST01191W1 | BEECH 1900D | Installation of Cargo Conversion Modifications. | | Installation in accordance with Raytheon Aircraft Services Master Drawing List RAS 0422-1, Revision N, dated July, 1 2003 or later approved revisions. Operation in accordance with Raytheon Aircraft Services Airplane Flight Manual Supplement Number 05 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 03.09.2008 |
| 10016551 | HAWKER BEECHCRAFT SERVICES | EASA.IM.A.S.02931 | REV. 1 | FAA STC ST00589LA | HAWKER 1000 | Installation of AlliedSignal Enhanced Ground Proximity Warning System (EGPWS). | | Raytheon Aircraft Services Master Data List for Allied Signal Enhanced Ground Proximity Warning System Installation on Hawker 100A Aircraft Report No. RAY-R3-002, Revision "A" dated April 3, 1998. To determine which configuration(s) or the EGPWS have b (...) | The installation should not be incorporated in any aircraft unless it is determined that the relationship between this installation and any previously approved configuration will not introduce any adverse effect upon the airworthiness of the Aircraft. (...) | 04.07.2008 |
| 10026878 | HAWKER BEECHCRAFT SERVICES | P-EASA.IM.A.S.03243 | | FAA STC SA01525W1 | BEECH 200, 200C, 200CT, 200T BEECH 300, 300LW, B200, B200C, BEECH B200CT, B200T, B300, BEECH B300C, HAWKER BEECHCRAFT B200GT, B200 | Installation of AirLock Secure Entry Door Lock System in place of original equipment locks for the main cabin door, nose/avionics doors, and aft equipment door. | FAR 23 | Installation in accordance with Hawker Beechcraft Services Master Data List Document NO 007-5200-L01-00 Revision 1, dated December 9, 2008, or later approved revisions. | None | 13.08.2009 |
| 10036781 | HAWKER BEECHCRAFT SERVICES | | | FAA STC SA00857W1 | BEECH 1900D | Installation of Universal Avionics System Corp UNS-1K Flight Management System. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Raytheon Aircraft Services Master Drawing List ref 148-100013, Revision B, dated February 24, 2000. Operation in accordance with Raytheon Aircraft Services Flight Manual Supplement ref FMS002048UE368, dated March 2, 2000 (...) | None. | 06.10.2011 |
| 10036782 | HAWKER BEECHCRAFT SERVICES | | | FAA STC SA00878W1 | BEECH 1900D | Installation of a forward double or triple club executive style interior configuration. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Raytheon Aircraft Services Master Drawing List ref RAS19D1000, Revision A, dated May 24, 2000. Operation in accordance with Raytheon Aircraft Services Flight Manual Supplement ref RAS19D10FMS, dated May 24, 2000 or later (...) | None. | 06.10.2011 |
| 10016046 | HAWKER BEECHCRAFT SERVICES, Inc. | EASA.IM.A.S.02217 | REV. 2 | FAA STC SA10478SC | BEECH B200, B200C BEECH B200GT, B200CGT BEECH B300, B300C | Installation of Digital Flight Data Recorder System | | Master Data List RAS060500131 Revision 12, 16 April, 2009 (>,<,>) or later approved revision. Airplane Flight Manual Supplement RAS060500132 Revision 2.0, dated 26 March, 2009 (>,<,>) or later approved revision. | None | 16.06.2009 |
| 10016686 | HEAD BALLOONS, INC. | EASA.IM.BA.S.01000 | | SL2263SO | HEAD BALLOONS AX7-77 HEAD BALLOONS AX7-77B HEAD BALLOONS AX8-105 HEAD BALLOONS AX8-88 HEAD BALLOONS AX8-88B HEAD BALLOONS AX9-118 | Installation of Aerostar (Raven) Models CW and RW baskets. | | FAA STC SL2263SO, Head Balloons Drawing No. R-1b, revised April 10, 1993; No. R2b, revised April 10, 1993; No. R-3a dated September 16, 1991; and No. R-4 dated April 10, 1993 or later FAA approved revisions. Flight Manual Supplements: Head AX7-77 Ballo (...) | The following Aerostar burners are approved for use with the above Head envelope / Aerostar basket configuration: Rally HP burner PN 13180HP II Dual burner PN 17395HP II Single burner PN 17398Rally HP II burner PN 17399HP III Dual burner part number 52350HP III Single burner PN 52370-01 | 13.04.2007 |
| 10016687 | HEAD BALLOONS, INC. | EASA.IM.BA.S.01001 | | FAA STC SB00463AT | HEAD BALLOONS AX7-77 HEAD BALLOONS AX7-77B HEAD BALLOONS AX8-105 HEAD BALLOONS AX8-88 HEAD BALLOONS AX8-88B HEAD BALLOONS AX9-118 | Installation of Cameron Balloons Ltd. baskets and burners underneath Head Balloons, Inc. envelopes | | Balloon Flight Manual (BFM) Supplements per model:- BFM supplement for Head Balloons AX7-77 w/Parachute top, revision 1, dated OCT 21 2008- BFM supplement for Head Balloons AX7-77b w/Parachute top, revision 1, dated OCT 21 2008- BFM supplement (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 19.11.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|-------------------|----------------|----------------------|--|---|---|---|---|--------------------------|
| 10015546 | HEADS UP TECHNOLOGIES, Inc. | EASA.IM.A.S.01437 | | FAA STC SA00558DE-D | DORNIER 328-100 DORNIER D328-100 MSN 3061 | Installation of a Pilot and Co-Pilot Digital Air Data Unit | | Master Data List, document No 34-00-0002-00, Revision IR, dated 23 July 2004 or later approved revision; Instructions for Continued Airworthiness, document 34-02-0002-00, Rev 4.0 dated 16 July 2004 or 34-02-0002-10 Rev 1.0 dated 28 June 2005, or later approved revision | 1. Airplane Flight Manual Supplement: AFMS Document 34-07-0002-00, Revision IR, dated 19 August 2004 or Document 34-07-0026-00, Revision IR or later approved revision 2. Revision IR or later approved revision 3. Revision IR or later approved revision 4. Revision IR or later approved revision 5. Revision IR or later approved revision 6. Revision IR or later approved revision 7. Revision IR or later approved revision 8. Revision IR or later approved revision 9. Revision IR or later approved revision 10. Revision IR or later approved revision 11. Revision IR or later approved revision 12. Revision IR or later approved revision 13. Revision IR or later approved revision 14. Revision IR or later approved revision 15. Revision IR or later approved revision 16. Revision IR or later approved revision 17. Revision IR or later approved revision 18. Revision IR or later approved revision 19. Revision IR or later approved revision 20. Revision IR or later approved revision 21. Revision IR or later approved revision 22. Revision IR or later approved revision 23. Revision IR or later approved revision 24. Revision IR or later approved revision 25. Revision IR or later approved revision 26. Revision IR or later approved revision 27. Revision IR or later approved revision 28. Revision IR or later approved revision 29. Revision IR or later approved revision 30. Revision IR or later approved revision 31. Revision IR or later approved revision 32. Revision IR or later approved revision 33. Revision IR or later approved revision 34. Revision IR or later approved revision 35. Revision IR or later approved revision 36. Revision IR or later approved revision 37. Revision IR or later approved revision 38. Revision IR or later approved revision 39. Revision IR or later approved revision 40. Revision IR or later approved revision 41. Revision IR or later approved revision 42. Revision IR or later approved revision 43. Revision IR or later approved revision 44. Revision IR or later approved revision 45. Revision IR or later approved revision 46. Revision IR or later approved revision 47. Revision IR or later approved revision 48. Revision IR or later approved revision 49. Revision IR or later approved revision 50. Revision IR or later approved revision 51. Revision IR or later approved revision 52. Revision IR or later approved revision 53. Revision IR or later approved revision 54. Revision IR or later approved revision 55. Revision IR or later approved revision 56. Revision IR or later approved revision 57. Revision IR or later approved revision 58. Revision IR or later approved revision 59. Revision IR or later approved revision 60. Revision IR or later approved revision 61. Revision IR or later approved revision 62. Revision IR or later approved revision 63. Revision IR or later approved revision 64. Revision IR or later approved revision 65. Revision IR or later approved revision 66. Revision IR or later approved revision 67. Revision IR or later approved revision 68. Revision IR or later approved revision 69. Revision IR or later approved revision 70. Revision IR or later approved revision 71. Revision IR or later approved revision 72. Revision IR or later approved revision 73. Revision IR or later approved revision 74. Revision IR or later approved revision 75. Revision IR or later approved revision 76. Revision IR or later approved revision 77. Revision IR or later approved revision 78. Revision IR or later approved revision 79. Revision IR or later approved revision 80. Revision IR or later approved revision 81. Revision IR or later approved revision 82. Revision IR or later approved revision 83. Revision IR or later approved revision 84. Revision IR or later approved revision 85. Revision IR or later approved revision 86. Revision IR or later approved revision 87. Revision IR or later approved revision 88. Revision IR or later approved revision 89. Revision IR or later approved revision 90. Revision IR or later approved revision 91. Revision IR or later approved revision 92. Revision IR or later approved revision 93. Revision IR or later approved revision 94. Revision IR or later approved revision 95. Revision IR or later approved revision 96. Revision IR or later approved revision 97. Revision IR or later approved revision 98. Revision IR or later approved revision 99. Revision IR or later approved revision 100. Revision IR or later approved revision | 15.08.2005 |
| 10015476 | HEATH TECNA, Inc. | EASA.IM.A.S.01229 | | FAA STC ST01532SE | B747-400 | Cabin Interior Reconfiguration | | Heath Tecna Inc documents for Project P861 & associated Master Drawing List HPD-DL-93 Rev A dated April 15, 2005 or later FAA approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 30.05.2005 |
| 10015574 | HEATH TECNA, Inc. | EASA.IM.A.S.01484 | | FAA STC ST00636SE | BOEING 737-300 BOEING 737-400 BOEING 737-500 | Installation of ATIX Interior retrofit Kit according to FAA STC ST00636SE | | Heath Tecna Drawing List HPD-DL-57, Revision F dated 17 January 2001 - FAA STC ST00363SE, Date reissued 17 March 2005 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 01.09.2005 |
| 10016255 | HEATH TECNA, Inc. | EASA.IM.A.S.02533 | | FAA STC ST01780SE | B747-400 | Installation of interior components in accordance with Heath Tecna Inc. Drawing List HPD-DL-105 | | Heath Tecna Inc. Drawing List HPD-DL-105, Revision B, dated July 25, 2007 FAA STC ST01780SE or later revisions of the above listed documents approved by EASA in accordance with ED Decision 2004/04/CF (or subsequent revisions of this decision). | Modification of the aircraft per Boeing Service Bulletin 747-00-2004, conversion from passenger to freighter configuration must be performed concurrently with this STC. | 29.08.2007 |
| 10016256 | HEATH TECNA, Inc. | EASA.IM.A.S.02534 | | FAA STC ST008700SE | B737-330 | Installation of galley supports structure and jumper wire provisions in accordance with Heath Tecna Drawing List HPD-DL-76, Revision B, dated 16, January 2004 or later approved revision. | | Health Tecna Drawing List HPD-DL-76, Revision B, dated 16, January 2001 or later approved revision. Weight & Balance Supplement WR-1027 SW, dated May 16, 2001 or later approved revision. | Approval of this change in type design applies to the abovementioned aircraft only. This approval should not be extended to aircraft on which other previously approved modifications are incorporated unless it is determined that the relationship between (...) | 24.10.2007 |
| 10016601 | HEATH TECNA, Inc. | EASA.IM.A.S.03015 | | FAA STC ST01154SE | B747-200B | Installation of interior components in accordance with Britax Heath Tecna Inc. (BHT) Drawing List HPD-DL-81 Revision B, dated December 12, 2002 or later FAA approved revisions | | Instructions for Continued Airworthiness Britax Heath Tecna Inc. HTM857 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 16.03.2009 |
| 10028538 | HEATH TECNA, Inc. | EASA.IM.A.S.03263 | | FAA STC ST01061SE | | LCD Monitor installation on Boeing Model 747-400 series aircraft in accordance with Heath Tecna, Inc. Master Drawing List HPD-DL-136, Revision E, dated May 5, 2010, or later FAA approved revisions. Validation of FAA STC ST02067SE | | ICA Supplement HTM957 Revision A, dated March 1, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 27.07.2009 16.07.2010 |
| 10030956 | HEATH TECNA, Inc. | | | FAA STC ST02067SE | B747-400 | LCD Monitor installation on Boeing Model 747-400 series aircraft in accordance with Heath Tecna, Inc. Master Drawing List HPD-DL-136, Revision E, dated May 5, 2010, or later FAA approved revisions. Validation of FAA STC ST02066SE | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. § 25.812(a)(1)(b)(1)(c)(i)(k)(l)(1)(2) Amdt 25-58 § 25.853(a)(d)(e) Amdt 25-58 § 25.869() | ICA Supplement HTM960 Revision E, dated May 20, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 19.07.2010 |
| 10030973 | HEATH TECNA, Inc. | | | FAA STC ST02066SE | B747-400 | Seat Installation and interior reconfiguration on Boeing Model 747-400 series aircraft in accordance with Heath Tecna, Inc. Master Drawing List HPD-DL-137, Revision B dated May 11, 2010, or later FAA approved revisions. Validation of FAA STC ST02066SE | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. § 25.812(a)(1) - FAR Amdt 25-58 § 25.853(a)(c)(d)(e) - FAR Amdt 25-58 § 25.869(a)(1)(4) - (...) | ICA Supplement HTM960 Revision E, dated May 20, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 19.07.2010 |
| 10035106 | HEATH TECNA, Inc. | | | FAA STC ST02072SE | A330-201, A330-202, A330-203 A330-223, A330-243 | Installation of New J Class Crew Rest for Qantas Airline A330-200 series. This STC installs a removable crew rest curtain and curtain header | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List, ref : HPD-DL-141 Revision E, dated 14 June 2010 Instruction for Continued Airworthiness, ref : HTM963, dated 14 May 2010 or later revisions of the above listed documents approved by EASA | none | 26.05.2011 |
| 10039535 | HEATH TECNA, Inc. | | | FAA ST02227SE | 747-400 | Installation of 2 new avionics racks in the EE bay of a Boeing 747-400 series aircraft in accordance with Heath Tecna Inc. Master Data List HPD-DL-162 Rev. A, dated April 12, 2012 or later approved revision. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. § 25.571(a)(b), ..., Amdt 25-86, ..., § 25.853(a), ..., Amdt 25-83 § 25.869(a)(4), ..., Amdt 25-113, (...) | Heath Tecna Inc. Master Data List HPD-DL-162 Rev. A, dated April 12, 2012 Instructions for Continued Airworthiness: Heath Tecna Inc. Doc HTM 1004, Revision B, dated April 4 (-> -<) 2012 or later revisions of the above listed documents approved by EASA in (...) | Approval of this change applies to Boeing 747-400 aircraft listed in HPD-DL-162 Rev. A, dated April 12, 2012 or later approved revisions only | 10.05.2012 |
| 10043471 | HEATH TECNA, Inc. | | REV. 1 | FAA STC ST02257SE | B777-200 | Title: Austrian Interior Reconfiguration. Description: Interior Reconfiguration as detailed in Project Specific Certification Plan SR1360-CP. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:- Special Condition(s): E-01, Installation of Inflatable Restraints- the following paragraph(s) at a later amendment: 25 (...) | Master Data List, Doc. No. HPD-DL-157, Rev. L, dated 20.12.2012; Instructions for continued airworthiness, Doc. No. HTM1012, Rev. A, dated 20.12.2012; or later revisions of the abovementioned documents approved by EASA in accordance with EASA ED Decision 20 (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 21.02.2013 |
| 10044049 | HEATH TECNA, Inc. | | REV. 3 | FAA STC ST02251SE | B767-300 | Austrian Interior Reconfiguration | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:- Special Condition(s): E-01, Installation of Inflatable Restraints- the following paragraph(s) at a later amendment: 25.3 (...) | Master Data List, Doc. No. HPD-DL-156, Rev. K, Dated 11.01.2013 ICA, Doc. No. HTM1010, rev B, Dated 10.12.2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 06.09.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|----------------------|----------------|----------------------|--|--|--|--|---|---------------|
| 10014040 | HEATHTECNA | EASA.A.S.01114 | | | B767-300 SERIES | Introduction of Stowage 4B and infl panel to British Airways aircraft as defined in the modification definition document and Declaration of Design and Performance (DDP) | | -Heath Tecna Modification document C935 and associated DDP dated 26/01/2005 or later approved revisions | | 11.02.2005 |
| 10014078 | HEATHTECNA | EASA.A.S.01170 | | | B747-400 | Reconfiguration of Cabin Interior from current 387SIA passenger cabin configuration (16F+58B+313Y for RT 514) to a 383 (12F+47B+324Y), Cathay Pacific Boeing 747-400 fleet standard passenger cabin configuration. | | Heath Tecna Modification document C933 and associated DDP dated 21/6/05 or later approved revisions | As defined in DDP, Aircraft Serial MSN 26548, Zero Occupancy. For this modification to be accepted on an EU registered aircraft, requires the operator to confirm that all the associate FAA and Hong Kong CDA modifications have been EASA approved. | 26.07.2005 |
| 10014143 | HEATHTECNA | EASA.A.S.01261 | | | B747-400 SN 25127 & 25128 | Heath Tecna Modification C943 - Cathay Interiors Modification. | | Heath Tecna Modification Document C943 & associated DDP dated August 1, 2005 or later approved revision. | 1. As defined in DDP Aircraft Serial Numbers 25172 & 25128.2. Zero Occupancy.3. For this modification to be accepted on an EU registered aircraft requires the operator to confirm that all the associated FAA & Hong King CAD modifications have been EASA approved. | 24.08.2005 |
| 10014298 | HEATHTECNA | EASA.A.S.01542 | | | B767-300 | Business Class Footstool Assemblies on British Airways Boeing 767-300 | | Heath Tecna Modification document C973 & associated DDP Issue 1 or later approved revision | This STC is limited to S/N 24340, 24341, 25204, 25444, 25442, 25732, 25826, 25828, 25829, 27140, 24335, 24336, 25831 and 25834. | 11.10.2005 |
| 10014519 | HEATHTECNA | EASA.A.S.02600 REV 1 | REV 1 | | B747-400 | Cathay Pacific B747 Cabin Re-Configuration | | Modification C1030 Revision 1 | 1. Restricted to MSN 27132, MSN 27069 & MSN 265522. Zero occupancy.3. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". (...) | 07.04.2008 |
| 10014998 | HEATHTECNA | EASA.A.S.03298 | | | DC-9 SERIES | Meridiana DC-9 Series Stow Bin Extension. | | Heath Tecna modification record C1037. | Restricted to ASN 49247, 49248, 49531, 49669, 49740, 49901, 49902, 49903, 53064, 49808, 53182, 49785, 49792, 49935, 49951, 49854, 53013. | 20.11.2007 |
| 10035635 | HELI AUSTRIA GmbH | | REV. 1 | | AS 355 F1, AS 355 F2 | Compliance of C20R installation with BCAR Section G and confirmation of ability of C20R to fly one engine inoperative profiles. This revision follows a transfer requested by Heli Austria GmbH. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. BCAR Section G The requirements for environmental protection and the associated certificate (...) | AS 355 F1 SUPP 11.2 - Take-off and Landing Procedures and Performance Data on Clear Airfield and Helipad with One Engine Inoperative AS 355 F2 SUPP 11.2 - Take-off and Landing Procedures and Performance Data on Clear Airfield and Helipad with One Engine (...) | none | 02.09.2013 |
| 10016868 | HELICOPTER TECHNOLOGY COMPANY | EASA.IM.R.S.01244 | | FAA STC SR01282LA | MD HELICOPTERS 369A MD HELICOPTERS 369D MD HELICOPTERS 369E MD HELICOPTERS 369FF MD HELICOPTERS 369H MD HELICOPTERS 369HE MD HELICOPTERS 369HM MD HELICOPTERS 369HS | Installation of a Tail Rotor Blade Assembly. (STC Classified as Basic by the FAA). | | Models 369A, H, HE, HM, HS; MDL-3000-01 rev N/C dated 7 April 2001 # Master Document List Models 369D, EMDL-3100-01 rev F dated 7 April 2001 # Master Document List Model 369FF MDL-3500-01 rev N/C dated 7 April 2001 # Master Document List | N/A | 15.03.2007 |
| 10015311 | HELICOPTERES GUIMBAL | EASA.E.S.01001 | REV. 2 | | O-360-J2A | Unleaded Fuel / Approval of alternative fuels for STC EASA.E.S.01001 | The Certification Basis for the STC EASA.E.S.01001 is applicable to this approval. CS-E, issue 1 dated 24 October 2003, for all affected areas, as identified in document Hélicoptères Guimbal No. J00-002 | Instructions for operation, maintenance and inspection:- Guimbal Installation, Operator's and Maintenance Manual J45-001, Issue 3 or later revisions of the above listed documents approved by EASA | None | 03.01.2013 |
| 10039707 | HELICOPTERS ITALIA S.R.L. UNIPERSON | | | | AS 365N2 | Installation of Audio Voice Warning on Radar Altimeter Thales AHV-9 (ref. Change HI-365-34-01). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Helicopters Italia - Rotorcraft Flight Manual Supplement HI365-34-01-FMS, Rev. 1, dated 16th January 2012 or later revisions of the above listed document approved by EASA - Helicopters Italia - Master Document List (MDL) HI-365-34-01-MDL, Rev. 2, dated 14th February 2012 - Helicopters Italia - Instal | Limited to S/N 6414 | 18.05.2012 |
| 10045424 | HELIFAB Inc. | | | FAA STC SR0485RC | BELL 212, 412, 412EP | Installation of Cabin Area Cargo Net, according to Master Drawing & Documents List HF-412-DL-101, Rev. L, in Bell 212, 412, 412EP Helicopters | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -HF-412-DL-101, Rev. L, Master Drawing & Documents List-HF-412-CCL-0003, Rev. IR, Compliance Checklist-HF-412-ST-0002, Rev. C, Load Test Plan (only Load Test results, Table 1)-HF-412-2550-01, Rev. E, Cargo Net Installation (Drawings) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 24.06.2013 |
| 10017144 | HELIKOPTER AIR TRANSPORT GmbH | EASA.R.S.01247 | | | EC 135 | Installation of Laser Scanning Equipment | | EO 85-001-AFM Supplement Doc. No. 9.2-85-001, Rev. 0 or later approved revision-Instruction for continued airworthiness Doc. No. EO 85-001 chapter 16, Rev. 0 or later approved revision-Installation manual Doc. No. IM 85-001, Rev. 0 or later approved revision | Limited to SN 025, 050, 053, 072, 097, 127, 128, 160, 168, 181, 182, 187, 196, 199, 205, 210, 220, 221, 305, 308, 325, 359, 368 and 387. | 01.06.2006 |
| 10027677 | HELIKOPTER AIR TRANSPORT GMBH | P-EASA.R.S.01499 | | | EC 135 T1 (CDS) EC 135 T2 (CPDS) EC 135 T2+ | Engineering Order EO 85-002, dated 21 September 2009 "Replacement of Siren Hooks with Onboard Talon Systems". | JAR 27 first issue - 06. September, JAR 27 Appendix C for Category A Operation, and JAR 27 Appendix B for operation under IFR; CS 27.1(a) in connection with CS 27.2(b)(2)(i) elect to comply ace. to CRI A-6. The Certification Basis for the original prod (...) | Installation Manual IM 85-002, dated 08.09.09; OnBoard CMM 528-023-01, Rev. 15, dated 23.04.09; Indraero Siren CMM SL07-3, dated 22.06.05; AFM Supplement 85-002 T1 (CDS), dated 08.07.09; AFM Supplement 85-002 T2(CPDS), dated 08.07.09; AFM Supplement 8 (...) | Installation does not constitute operational approval. Approval of PCDS (Personal Carrying Device System) for external load operation must be granted by the competent authority. | 20.10.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|--------------------------|---|--|---|---|--|---------------|
| 10033490 | HELIKOPTER AIR TRANSPORT GmbH | | | | EC 135 T1(CDS) EC 135 T2(CPDS), EC 135 T2+ | Replacement of Onboard Talon System with improved IAOSI hook AS 21-17 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Engineering Order EO 85-004 dated 11. October 2010- Instructions for continued airworthiness, Chapter 11 of EO 85-004 dated 11. October 2010; - AFM Supplement 9.2-85-004 for EC 135 T1 (CDS); - AFM Supplement 9.2-85-004 for EC 135 T2(CPDS); - AFM Supple (...) | Installation does not constitute operational approval. Approval of PCDS(Personal Carrying Device System) for external load operation must be granted by the competent authority. | 25.01.2011 |
| 10034657 | HELIKOPTER AIR TRANSPORT GmbH | | | | EC135 T2 | Dual GNS430W Installation in Autopilot helicopter | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AFMS 34-002, Revision 0 or later revisions of the above listed documents approved by EASA Engineering Order EO 34-002, Revision 0, including referenced drawings and Appendix A and Installation Manual EO 34-002 CMM 34-002 Maintenance Instructions | The installation is limited to Aircraft Serial Numbers 0220 and 0221 | 18.04.2011 |
| 10016751 | HELI-LYNX | EASA.IM.R.S.01078 | | TCCA STC SH05-8, ISSUE 1 | EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 | Installation of Anticipator of Governor Cables Ass. Refer to installation manual IM, PA 04 015 91 01 issue 1 or later approved | | Refer to installation manual IM, PA 04 015 91 01 issue 1 or later approved. | 1. Refer to instructions for continued airworthiness ICAPA 04 015 05 01 issue 1 or later approved and associated maintenance instructions. 2. This STC is approved only for the product configuration as defined in the approved design data referre (...) | 10.10.2005 |
| 10016752 | HELI-LYNX | EASA.IM.R.S.01079 | | SH04-27 | EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 | Heli-Lynx AS 355 FX System modifications, as per Transport Canada STC n° SH04-27 | | Heli-Lynx Helicopter Services Master Drawing MDL-PA-03-007-91-01, Issue 19, TCCA approved September 27th, 2004, or later TCCA approved revisions. Installation Heli-Lynx Helicopter Services Installation Manual IM-PA-03-007-91-01, Issue 3, revision C, T (...) | | 04.04.2007 |
| 10016818 | HELI-LYNX | EASA.IM.R.S.01183 | | SH05-41 | EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 | Installation of One or Two Concorde RG 355 Laed-Acid Batteries according to TCCA STC SH05-41 issue N°2 approved September 09, 2005 | | Installation Manual IM-PA-04-021-91-01 issue 1 Rev A, Flight Manual Supplement FMS-PA-04-021-91-01 Issue 1 Rev A, Instruction for continued airworthiness ICA-PA-04-021-05-01 Issue 1 Rev A, all approved 09 September 2005 or later approved | Refer to STC SH05-41 and Flight Manual Supplement | 16.03.2006 |
| 10016819 | HELI-LYNX | EASA.IM.R.S.01184 | | SH05-40 | EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 | Installation of Purolator De-icing Fuel Filter p/n 1743640-01 | | Heli-Lynx Helicopter Services Master Drawing List n° MDL-PA-04-012-91-01 Issue 1, Rev A, approved by Transport Canada (TCCA) on September 3rd, 2005, or later TCCA approved revisions. Installation Manual Report n° IMPA-04-012-91-01 Issue 2, Rev. A, (...) | This STC is the EASA validation of Transport Canada STC n° SH05-40, issue 1, dated September 9', 2005, | 25.04.2008 |
| 10016938 | HELI-LYNX | EASA.IM.R.S.01326 | | TCCA STC SH05-20 ISSUE 2 | EUROCOPTER FRANCE AS 350 EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 C EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 EUROCOPTER FRANCE AS 350B EUROCOPTER FRANCE AS 355 SERIE EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 EUROCOPTER FRANCE AS 355N | AS350/AS355 Rear Shoulder Harness Installation. (Modification approved by TCCA under STC SH05-20 Issue 2.) | Definition and Installation: Master Drawing List No. MDL-PA-04-014-91-01, issue 6, rev. A. Installation Manual No. IM-PA-04-014-91-01, issue 4, rev. A. Inspection and Maintenance: Instructions for Continued Airworthiness No. ICA-PA-04-014-05-01, issue 2, rev. B. | Installation of a Rear Shoulder Harness on AS350B3 with the modification OP-3369 incorporated is not approved. | 04.09.2007 | |
| 10035276 | HELI-ONE | | | TCCA STC SH05-46 | S-61N | Sealed Lead Acid Battery Installation | CAR-7 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and (...) | Flight Manual Supplement CHS61-24-016 Revision A or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision) CHC Approval Control Document CHS61-24-016 Revis (...) | none | 15.06.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|-------------------|----------------|--------------------------|--|---|---|---|--|---------------|
| 10027687 | HELI-ONE (NORWAY) AS | P-EASA.R.S.01307 | | | AS332C AS332C1 AS332L AS332L1 | EFIS 520 Installation. | FAR 29. | AHS-028 original issue dated 14 July 2009 or later approved revision - Flight Manual Supplement; 52200 revision H dated 9 March 2009 - Master Drawing; AHSE-54559 Original issue dated 10 October 2009 - Instructions for Continued Airworthiness; AHSB-3420-055 original issue dated 6 December 2008 - Service B | None. | 21.10.2009 |
| 10036393 | HELI-ONE (NORWAY) AS | | | | AS332 L1/L2 | Installation of STAR SAFIRE FLIR system consisting of external mount, internal and external cabling and installation of mission console in Cabin. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AHS-124 dated 24 May 2011 - Flight Manual Supplement FLIR STAR SAPHIRE HD or later revisions of the above listed documents approved by EASA; AHSE-56756 revision C dated 30 August 2011 - Compliance Checklist; AHSE-57366 revision A dated 24 May 2011 - Con (...) | Flight into known Icing Conditions prohibited. | 06.09.2011 |
| 10040394 | HELI-ONE (NORWAY) AS | | | | AS 332 L, AS 332 L1 | Goodrich Dual Hoist Installation | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AFM Supplement AHS-090, dated 07.04.2011 Continued Airworthiness Supplement No: AHSE-55211, Rev: B, dated 12.01.2012 or later revisions of the above listed documents approved by EASA; Master Drawing AHSE-55200, Rev: B, dated 20.06.2012 | None | 29.06.2012 |
| 10037034 | HELI-ONE CANADA INC. | | | | S-61L & N | Lower Housing Weld Repair | The Certification Basis for the original product remains applicable to this certificate/ approval. CAR 7 & FAR Part 29 (As per FAA TCDS 1H15) The requirements for environmental protection and the associated certified noise and/or emissions levels of (...) | Helix-One Engineering Order, E0 No: H1E-021 Rev. 0, Transport Canada Approved 12 April 2007. | Applicable only to Sikorsky S-61N Lower Housing Assembly Part Nos S6135-20670-9, S6135-20670-42, S6135-20670-43, S6135-20629-102, S6135-20629-104. | 26.10.2011 |
| 10016796 | HELI-ONE NORWAY AS | EASA.IM.R.S.01149 | | | AS 332 L1 | BPOL 332L1 Police equipment Installation | | AHS-005 09/05/2006 TDR-94D ATC Transponder 332L1 BGSAHS-027 08/05/2006 Aerodata Aerophone Iridium Installation 332L1 BGSEASA-approved Flight Manual supplements AHS-053 12/10/2006 AeroNav II with GPS 332L1 Bp0AHS-054 20/04/2006 NV (...) | Helicopter modified in accordance with this STC are not approved for NVG operations | 12.06.2006 |
| 10040010 | HELI-ONE NORWAY AS | | REV. 3 | | EC225 LP | Installation of LH or RH or both Vertical Reference Door. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | SUP.H1No 165G dated 31 May 2013 - Flight Manual Supplement or later revisions of the above listed documents approved by EASA; AHSE-62232 revision D dated 11 April 2013 - Certification Programme AHSE-62237 revision A dated 4 May 2012 - Continued Airworth (...) | Day/Night VFR Only; Night VFR flight limited to normal transfer/ferry flights only, no vertical load operations permitted; Flight into known icing conditions prohibited; Category A operations prohibited; Operations requiring certification for Ditching p (...) | 04.06.2013 |
| 10043033 | HELI-ONE NORWAY AS | EASA.R.S.01306 | | | AS 365 N2/N3 SA 366 N/N1 SA 366 G1 | Installation of Penny & Giles D51615 MPFR CVFDR. | | - Definition: Helix-One Master Drawing ref. AHSE-53500 revision A, dated 13 January 2007, or later EASA approved revision - Installation Engineering Order ref. EOM-3130-26597, Revision A dated 21/08/06, or later revision - Instructions for continued air (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined (...) | 09.03.2007 |
| 10043034 | HELI-ONE NORWAY AS | EASA.R.S.01581 | | CAA UK AAN 25867 ISSUE 2 | SIKORSKY S-61N | Installation of Automatically Deployable Emergency Locator Transmitter (ADELT) | | - Bond Modification No. 1731- Flight Manual Bond Supplement No. 5 or later approved revisions Note: This STC reflects the transfer of CAA UK AAN 25867 Issue 2 from Bond Helicopters Ltd. to Helix-One Norway AS | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product., This C (...) | 09.07.2009 |
| 10043037 | HELI-ONE NORWAY AS | P-EASA.R.S.01347 | | | SIKORSKY S-61N | AHSE-53880 Breeze 29700 Stand by Rescue hoist installation. | | - RFMS AHS-077 dated 26.06.2008 or later approved revision - Master Drawing AHSE 53880 28.08.2008 or later approved revision - Continued Airworthiness Supplement AHSE-53888 Rev A dated 20.06.2008 or later approved revision. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The in (...) | 24.09.2008 |
| 10017036 | HELI-ONE USA | EASA.IM.R.S.01460 | | FAA STC SH8601SW | BELL 230 | Installation of ED2-705 EFIS four tube flight instrument System. (This STC was classified as Basic by the FAA and is approved using the Post Type Validation Principles.) | | HD230-00-0002 rev I/R dated 22 December 1992 or later approved revision # Drawing List; HD2-92-91 dated 8 February 1993 or later approved revision # Flight Manual Supplement | | 13.05.2008 |
| 10017042 | HELI-ONE USA | EASA.IM.R.S.01466 | | FAA STC SH5880SW REV2 | BELL 230 | Installation of SPZ-7000 DAFCS System. | | HD230-00-0001 revision I/R dated 22 December 1992 or later approved revision # Helix-Dyne Drawing List; HD2-92-24 dated 8 February 1993 or later approved revision # Flight Manual Supplement. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 05.06.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|----------------------|--|--|---|--|---|---------------|
| 10043049 | HELIPOD USA LLC | EASA.IM.R.S.01213 | | FAA STC SR00038BO | R22 | Installation of externally mounted cargo carrying pods in accordance with FAA STC SR00038BO Issued January 21, 1995, reissued January 28, 2004 and amended May 25, 2006. | | Installation of Helipod USA containers in accordance with Helipod USA Master Drawing list HSL-A-027-MDL, rev. 1, dated March 31, 2005 or later FAA approved revision. Weight and balance for the installation of 18 pounds per Helipod at longitudinal arm of (...) | 1. Helipod USA pilot's operating Handbook Supplement No1, rev A dated 23 March 2005, for Robinson R22 helicopters is required and must be carried during all flights. The approval should not be extended to other aircraft of these models or which other (...) | 26.06.2006 |
| 10017093 | HELISURESTE | EASA.R.S.01005 | REV. 1 | | BO105 CB-4, BO105 CBS-5 | Installation of FLIR Ultra force II System The existing FLIR Ultra 400 system installation is adapted to use the FLIR Ultra Force II system. UF2 platform, adapter plate, CEU, CEU mounting tray, LCU and power supply panel are new installed. Bolster, etc (...) | | - Doc No. CMA021 "Documento Técnico", Ed. 1, dated 27 July 2004- Doc No. CMA021-ID "Informe sobre el Diseño", Ed. 1 Installation Instructions in accordance with:- Doc No. CMA021-ID "Boletín de Ingeniería", Edición 1- Doc No. CMA021-MID "Instrucciones (...) | Subjected to the conditions, limitations and procedures contained in Doc No. CMA021-AFMS "Supplement for FLIR System - Ultra Force II", Iss 1 Rev. 0 | 28.02.2007 |
| 10017113 | HELISURESTE | EASA.R.S.01090 | | | AGUSTA BELL 412 BELL 212 BELL 412 | Installation of a pilot heater failure warning indicator | | Associated Technical Documentation listed on Technical Document CMA077 Issue 1 Rev. 0 or later approved revisions. Installation / Instructions for Continued Airworthiness in accordance with: # Engineering Bulletin CMA077-BI Issue 1 Rev. 0 or later a (...) | | 09.07.2007 |
| 10017114 | HELISURESTE | EASA.R.S.01094 | | | AB-412, BELL 212, 412, 412EP | Installation of a standby horizon emergency power supply warning indicator. | | Associated Technical Documentation listed on Technical Document CMA078 Issue 1 Rev. 0 or later approved revisions. Installation / Instructions for Continued Airworthiness in accordance with: # Engineering Bulletin CMA078-BI Issue 1 Rev. 0 or later a (...) | | 09.07.2007 |
| 10017152 | HELISURESTE | EASA.R.S.01255 | | | SA 330 J | Installation of an external basket and external reinforcement of the structure (as necessary) to carry a Bambi Bucket and its accessories during fire fighting operations. | | Flight manual Supplement CMA098-AFMS/1 Issue 1 Revision 0 dated 6 February 2008 or later approved revision. Maintenance & Installation/Removal instructions CMA098-MID issue 1 revision 0 or later approved revision. Technical Document CMA098 Issue 1 revision 0 or later approved revision. | N/A | 08.02.2008 |
| 10017224 | HELISURESTE | EASA.R.S.01389 | | | BELL 412 BELL 412 EP | CMA 135 "Radio Altimeter Audio Warning Indicator". | | Technical Document CMA 135 Issue 1 or later approved revision. Installation instructions for continued airworthiness in accordance with Engineering Bulletin CMA 135-BI Issue 1 or later approved revisions, Drawing CMA135-02-00-00 or later approved rev (...) | | 01.02.2008 |
| 10017118 | HELISURESTE CENTRO DE MANTENIMIENTO | EASA.R.S.01125 | | | SIKORSKY S-61N | External hoist P/N 76378-260-D replacement by P/N 76378-500 | | CMA095-AFMS Supplement for Air Equipment Electric Hoist Goodrich P/N 76378-500. CMA095-MID Maintenance and Installation/Removal Manual. | Sikorsky S-61N operated operated by HELICSA (S/N 61756, 61758, 61299, 61361 & 61741). | 16.10.2007 |
| 10043039 | HELISURESTE CENTRO DE MANTENIMIENTO | P-EASA.R.S.01093 | | | A 109E | Installation of an image capture system (ARGUS 275 Camera) by adaptations of the previous installation of a SX-16 searchlight & transmission system. | | - CMA015-AFMS Supplement for an Image Capture System (Argus275) and Transmission System- CMA015-BI Engineering Bulletin- CMA015-MID Installation/Maintenance Manual | 1. The installation of the supporting structure (P/N 109-0709-21-101 and P/N 109-0709-24-101) and reinforcements (P/N 109-0709-66-101, P/N1090709-23-109, P/N 109-0709-23-113, and P/N 109-0858-65-113) related to the Agusta optional equipment searchlight (...) | 17.06.2009 |
| 10014012 | HELISURESTE CMA | EASA.A.S.01055 | | | FTB-337-G | Change to the complete avionics set installed by a new one last generation GARMIN and HONEYWELL | | "Documento Técnico CMA014, Issue 1" (Aircraft Flight Manual Supplement "CMA014-AFMS, Issue 1" included) | Restricted to: s/n FTB3370020, FTB3370021, FTB3370022, FTB3370023, FTB3370024 & FTB3370025 | 14.09.2006 |
| 10016843 | HELI-TRADE CORPORATION | EASA.IM.R.S.01219 | | SR01592SE | BELL 205A-1 BELL 205B BELL 212 BELL 214ST BELL 412 | Installation and Maintenance of Goodrich Internal Hoist FAA STC SR01592SE | | Master Drawing List HI-415-MDL dated August 10, 2005 or later approval revision; RFM Supplement FMS-415-29-1581 dated August 12, 2005 or later approved revision | In order to operate the internal hoist in the referenced airframes, the following provision kits must be installed: HOIST P/N BELL PROVISION KIT BELL SERVICE INSTRUCTION HELICOPTER S/N EFFECTIVITY HTC-42305A 205-706-091-1 BHT-205-SI-44 205A-1 30001-30229HTC-42305A 205-70 | 03.10.2006 |
| 10017279 | HELI-UNION | EASA.R.S.01525 | | | AS 365 N | GPS Stand alone BRNAV FREEFLIGHT TNL 2001. Approach plus I/O Installation | | 2 Dossier de Modification M003-34-08-365N Révision 2; Héli-Union Supplément Manuel de Vol BRNAV SA 365N, 365HU-PMV-34-07 (08-44). | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 16.03.2009 |
| 10043536 | HELI-UNION | | | | MBB-BK117 C-2 | Replacement of the FLIR analogue video recorder SONY DSR-20MDP by digital video recorder SKYQUEST VDRV-3000-01, as detailed in M004-85-11-145. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Dossier de Modification M004-85-11-145 Rev. 2; Installation instructions and Instructions for Continued Airworthiness: Service Bulletin "Installation VRDV EC145", 145HU-SB-85-68 Rev. 1; or later revisions of the above listed documents. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 04.02.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|----------------------|--|---|---|---|--|---------------|
| 10046740 | HEL-UNION | | | | MBB-BK117 C-2 | UHF TPM700 Installation. Installation of a UHF TPM700 radio as defined in Dossier de Definition M001-23-13-145. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Dossier de Modification M001-23-13-145 Ed. 1, Rev. 2, dated 30 September 2013. Installation Instructions and Instructions for Continued Airworthiness, 145HU-SB-23-69 Rev. 1, dated 30 September 2013 | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 16.10.2013 |
| 10016727 | HERITAGE AVIATION LIMITED | EASA.IM.R.S.01028 | | FAA STC SR09407RC | EUROCOPTER FRANCE EC 155 B1 | Installation of Forward and Aft Facing Passenger Couch | | Heritage Aviation Master Drawing List 700-13000 Revision _ dated April 29, 2004 or later approved revision. Instructions for continued Airworthiness for Three/Four Place Passenger Couch Installation, Heritage Aviation Ltd. Document No. 708-73000 dated April 2, 2004 or later accepted revision. | | 20.07.2005 |
| 10016768 | HERITAGE AVIATION LIMITED | EASA.IM.R.S.01099 | | FAA STC SR00530NY | EUROCOPTER FRANCE EC 155 B1 | Installation of a Cocoon wall panel system | | Advanced Flight Systems Inc. Installation Drawing List Report No. 913-030-001 Original Revision dated 15 June 2001 or later approved revisions. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.07.2005 |
| 10017029 | HERITAGE AVIATION LIMITED | EASA.IM.R.S.01448 | | FAA STC SH7733SW | MD HELICOPTERS 369E MD HELICOPTERS 369FF MD HELICOPTERS 500N MD HELICOPTERS 600N | Installation of enlarged instrument panel and slant console. This STC has been classified as #BASIC# by the FAA and has been approved under the PTVF. | | A00-13003 revision B dated 14 June 2007 or later approved revision # Master Drawing List: A09-63001 revision Original dated 2 July 2007 or later approved revision # Flight manual Supplement: A04-73003 revision Original dated 22 March 2007 or later appr (...) | Not compatible with the MDHI optional centre seat installation. | 01.04.2008 |
| 10017048 | HERITAGE AVIATION LIMITED | EASA.IM.R.S.01472 | | FAA STC SR09483RC | MD HELICOPTERS 369E MD HELICOPTERS 369FF MD HELICOPTERS 500N MD HELICOPTERS 600N | Heritage Aviation, Ltd. STC SR09483RC; Replacement Instrument Panel (including pedestal, shrouds and glare shield) on applicable MDHI 369E, 369FF, 500N & 600N helicopters. | | - Compliance Checklist; E07-1241 Rev.A or subsequent approved revision.- Master Drawing List; A00-13001 Rev.C or subsequent approved revision.- Instructions for Cont.A/w; A04-73001 Rev.- (26.4.07) or subsequent approved revision. | 1. Applicable to MDHI 369E, 369FF, 500N and 600N helicopters.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will i (...) | 19.03.2009 |
| 10017049 | HERITAGE AVIATION LIMITED | EASA.IM.R.S.01473 | | FAA STC SR09484RC | MD HELICOPTERS 369E MD HELICOPTERS 369FF MD HELICOPTERS 500N | Heritage Aviation, Ltd. STC SR09484RC; Chelton EFIS installation on applicable MDHI 369E, 369FF & 500N helicopters. | | - Compliance Checklist; E07-1242 Rev.A or subsequent approved revision.- Master Drawing List; A00-14002 Rev.D or subsequent approved revision.- Instructions for Cont.A/w; A04-74002 Rev.- (7.5.07) or subsequent approved revision.- RFM Supplement; A09-54004 Revision B or subsequent approve | 1. Applicable to MDHI 369E, 369FF, 500N.2. Operation with the Chelton EFIS is limited to VFR only.3. (See limitations specified in RFM Supplement)4. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce (...) | 19.03.2009 |
| 10017068 | HERITAGE AVIATION LIMITED | EASA.IM.R.S.01504 | | FAA STC SR09481RC | MD HELICOPTERS 369E MD HELICOPTERS 369F MD HELICOPTERS 369FF MD HELICOPTERS 500N MD HELICOPTERS 600N | Heritage Aviation, Ltd. STC SR09481RC; Installation of noseshelf on applicable MDHI 369E, 369F, 369FF, 500N & 600N helicopters. | | Compliance Checklist; E07-1240 Rev.A or subsequent approved revision. Master Drawing List; A00-13002 Rev.B or subsequent approved revision. Instructions for Cont.A/w; A04-73002 Rev.- (26.4.07) or subsequent approved revision. | 1. Applicable to MDHI 369E, 369F, 369FF, 500N and 600N.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce (...) | 19.03.2009 |
| 10016813 | HERTELENDY RESEARCH ASSOCIATES | EASA.IM.R.S.01174 | | FAA STC SR09066RC | 47E, 47G, 47G-2, 47H-1, 47J... BELL 47, 47B, 47B3, 47D, 47D1. | Bell 47 Composite Tail Rotor. | | 1. Master Drawing and Document List QA.001-47-400-2 Rev.19 dated 1 April 2008;2. Installation and Maintenance Manual; QA.001-47-400-1 Rev. 6 dated 1 April 2008;3. HAC Flight Manual Supplement QA-FM-SUB-REQ-FAA-1-10-06-004 - Tail Rotor Pr (...) | 1. Bell 47 Series Rotorcraft equipped with Bell 47-641-170 hub and associated Bell 47-640-075 gear box only;2. ECN 07-07 dated 16 July 2007 'Installation of lighting protection grounding straps' must be installed;3. Blade life 4000 hours.4. (...) | 13.05.2008 |
| 10016082 | HOLLINGSEAD INTERNATIONAL | EASA.IM.A.S.02266 | | FAA STC ST01879LA | BOEING 747-400 SERIES | Installation of B/E Aerospace Air Chiller P/N 470-1 | | B/E Aerospace Master Data List EC470-9, Revision C, dated November 6, 2006, or later approved revisions | This approval is limited to airplanes on which C&D Zodiac Pantry P/N C481001-101 has been installed in accordance with STC ST01880LA | 23.01.2007 |
| 10016189 | HOLLINGSEAD INTERNATIONAL | EASA.IM.A.S.02418 | | FAA STC ST00740LA-D | BOEING 737-300 | Installation of Smoke Detection and Fire Suppression Systems Provisions, Lower Cargo Compartments. | | Hollingshead International, Inc. Drawing No. 90446 (-506/-507) Rev.A dated 5/19/99 and Drawing List 12114 Revision A dated 5, 24/99 or later EASA approved revisions. | | 09.10.2007 |
| 10016262 | HOLLINGSEAD INTERNATIONAL | EASA.IM.A.S.02541 | | FAA STC ST00404LA-D | BOEING 737-400 | Installation of Smoke Detection and Fire Suppression Systems Provisions, Lower Cargo Compartments. | | Drawing No 90437 (-508/-507) Rev.N/C dated 03/23/99, and MDL 12111 Rev. G, dated 15 May 2002; Drawing No. 90437 (-508&-509) Rev. A dated 10/15/99, and MDL 12111 Rev. D, dated 05/02/00; FAA approved Aircraft Flight Manual Supplement 20673 Rev.B dated 02 (...) | | 14.01.2008 |
| 10016300 | HOLLINGSEAD INTERNATIONAL | EASA.IM.A.S.02597 | | FAA STC ST02147LA | BOEING 777-200 SERIES | Installation of B/E Aerospace air Chiller P/N 470-1 | | FAA Letter 150L-08-184 dated June 17, 2008. | Applicable only to A/C with C&D Zodiac pantry installation P/N C801001-101 and C801001-103 covered by FAA STC ST02144L and EASA STC P-EASA.A.S.02421. | 18.06.2008 |
| 10016499 | HOLLINGSEAD INTERNATIONAL | EASA.IM.A.S.02861 | | FAA STC 01835LA | BOEING 767-200 BOEING 767-300 | Installation of mounting provisions and electrical provisions for electronic flight bags. | | Hollingshead International Master Data List 2813-01, Revision D, February 20, 2007 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 02.06.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|-------------------|----------------|------------------------|---|--|---------------------|--|---|---------------|
| 10015343 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.00442 | REV 3 | FAA STC ST00705SE | BOEING 747-400, 747-200 | Installation of the Enhanced Ground Proximity Warning System on a Boeing 747-200/400 aircraft. | | Master Drawing, Enhanced Ground Proximity Warning System Boeing 747 Series Aircraft, Honeywell document 060-4199-094, rev. BC. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. For Boeing B747-... | 29.01.2007 |
| 10015348 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.00498 | | FAA STC ST01124SE | BOEING 747-400 SERIES | Installation of the Flight Data Acquisition Management System | | - Master Drawing List 756-0082-001, rev A, or later FAA approved revisions. - FAA STC ST01124SE | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer | 10.01.2005 |
| 10015371 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.01014 | REV 1 | | SEE STC LIST | Installation of Honeywell TFE731-3D-1H or TFE731-3DR-1H Turbofan Engines. | | 1. Garrett Aviation Services drawing list DL456 dated February 14, 1996.2. AFM Supplement Landmark Aviation document 1013496, Rev. (A), dated August 9, 2007.3. Master Data List DL456, dated July 10, 2007, revision (C). | | 31.03.2008 |
| 10015377 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.01051 | REV 1 | FAA STC NO ST02005CH-D | LEARJET 55 | Installation of Honeywell N1 Digital Electronic Engine Control (DEEC)Note: This revision reflects the transfer of STC EASA.IM.A.S.01051 from Garrett Aviation Services to HoneywellInternational Inc. | | -Garrett Aviation Services Drawing List 1001811 Revision A dated October 5 2004 or later approved revision-AFM Supplement 1002971 Revision A dated October 7 2004 or later approvedrevision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 02.04.2009 |
| 10015462 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.01209 | | FAA STC ST01298AT | BOEING 767-200 SERIES BOEING 767-300 SERIES | Installation of Flight Data Acquisition & Management System (FDAMS) with Mark II Aircraft Communication & Reporting System (ACARS) | | Honeywell International Master Drawing List 699-50117 Revision K dated August 6 2001 or later approved revision. | 1. Pitch & Roll Attitude is sampled one per second.2. Pneumatic power is sampled once every 4 seconds. | 30.06.2006 |
| 10015491 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.01253 | | FAA STC ST00961AT | BOEING 757-200 SERIES | Installation of Flight Data Acquisition and Management System (FDAMS), FAA STC ST00961AT. | | Honeywell International Inc. Master Drawing List for the Installation of the Honeywell FDAMS in the Boeing 757-200 Series Aircraft,document number 699-50109, Revision K or later FAA approved revision. | | 29.06.2005 |
| 10015547 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.01438 | | FAA STC SA3122SO | BOEING 747 | Honeywell International Inc. of Redmond, Washington, USA has applied for an STC to cover the installation of a CASR1 Traffic Collision Avoidance System (TCAS) II with dual TRA-67 Mode S transponders on Boeing 747-100/200/SP Series aircraft. The FAA has issued STCSA3122SO for this modification. | | Refer to Approval Report IVW/LTB/TCP-05.533054 | 1. Refer to Approval Report IVW/LTB/TCP-05.533054. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 17.08.2005 |
| 10015572 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.01479 | | FAA STC SA00842WI-D | BEECH 1900D | Installation of the Honeywell Mark VI Enhanced Ground Proximity Warning System | | Master Drawing List 159-08250-000 1 Rev. A dated 15 December 1999Airplane Flight Manual Supplement 006-00875-0000, Rev.A dated April 2001 | AFMS 006-00875-0000 is required on-board for operation withBeech 1900 D series | 28.11.2005 |
| 10015575 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.01485 | | FAA STC SA2726SO | DC-9-80 (MD-80) SERIES | Installation of CAS-81 TCAS II SystemFAA STC SA2726SO | | Allied Signal Master Drawing List No. 699-50005 Rev. M dated 4 June 2003.Aircraft Flight Manual Supplement No. 699-52554 Rev. C dated October 1999. | | 02.05.2006 |
| 10015685 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.01670 | | FAA STC SA01116WI-D | PILATUS PC-12/45 | Single or Dual Altimeters (FAA STC SA01116WI-D) | | For Single Altimeter: PC-12, PC-12/45 only MDL No. 159-08290-0001 Rev. B, dated 18/10/2002For Dual Altimeter: PC-12, PC-12/451. MDL 159-08290-0003 Rev. B, dated 20/02/20042. AFMS No. 006-10318-0000, dated 20/02/2004For Dual Altimeters: PC-12/471. MD (..) | PC-12/45: MSN 101 to 835 (except MSN 645)PC-12/47: MSN 684 to 835 | 15.05.2007 |
| 10015686 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.01671 | | FAA STC SA01253WI-D | PILATUS PC-12/45 | RYSM (FAA STC SA01253WI-D issue dated 20/02/2007) | | PC-12/45:MDL No. 159-08297-0001, Rev. A, dated 28/05/2004.AFMS No. 006-10319-0000, dated 31/08/2004.PC-12/47:MDL No. 159-08297-0001, Rev. E, dated 11/01/2006AFMS No. 006-10319-0000, Rev. D, dated 27/02/2006Later approved revisions to the above list (..) | | 15.05.2007 |
| 10015745 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.01776 | | FAA STC ST00491SE | BOEING DC-9-81 (MD-81) BOEING DC-9-82 (MD-82) BOEING DC-9-83 (MD-83) BOEING DC-9-87 (MD-87) BOEING MD-90-30 | FAA STC ST00491SE "Installation of the Enhanced Ground Proximity Warning System" | | Master Drawing List 960-2227-007 Rev. AY dated 7 June 2005AFM as indicated in Master Drawing List 960-2227-007 Rev. AY dated 7 June 2005 | | 05.03.2007 |
| 10015756 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.01790 | | ST01619SE | BOEING 747-400 | Introduction of Honeywell International Inc. Main/Nose Wheel pn. 2615901-1 and Carbon Brake pn. 2615902-1 in accordancewith Honeywell Service Bulletin 2615901-32-001_2615902-32-001 Revision New or later FAA-approved revision as replacement for Goodr (..) | | See item 5 of approval report (attachment) | | 22.03.2006 |
| 10015805 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.01870 | | FAA STC ST00705SE | BOEING 747-100 BOEING 747-100F BOEING 747-300 BOEING 747-SP | Installation of Honeywell Enhanced Ground Proximity WarningSystem (EGPWS). | | Master drawing 060-4199-094, revision BF, dated 13-11-2006 or later revisions of the above listed documents approved byEASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | | 06.09.2007 |
| 10015806 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.01871 | | FAA STC ST00140AT | AIRBUS A310-200 AIRBUS A310-300 | Installation of MUA-45 Data Link System in Airbus A310-200 / -300 series aircraft. | | Master Drawing List, ref: 699-50053, Revision B, dated 05 September 1995. | | 12.04.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|-------------------|----------------|----------------------|--|---|---------------------|--|---|---------------|
| 10015847 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.01947 | | FAA STC ST00417SE | BOEING 767-200 BOEING 767-300 BOEING 767-300F | Installation of Enhanced Ground Proximity Warning System (EGPWS) | | Master Drawing List 960-222 Rev N dated June 2, 2002 or later approved revision | Honeywell International Inc. Airplane Flight Manual Supplements 060-4189-003 Revision B, 060-4189-014 Revision A, 060-4189-008 Revision A, 060-4189-037 Initial Revision, 060-4189-052 Revision A, 060-4189-013 Initial Revision, 060-4189-068 Revision A (...) | 07.04.2006 |
| 10015893 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.02013 | | FAA STC ST00446SE | DC-9-80 (MD-80) SERIES | Installation of the Honeywell Solid State Flight Data Recorder | | Master Drawing 755-3026-001 Rev. C dated 9-3-2006 | -- | 11.09.2006 |
| 10015921 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.02054 | | FAA STC ST00389SE | BOEING 737-200 BOEING 737-300 BOEING 737-400 BOEING 737-500 BOEING 737-700 BOEING 737-800 BOEING 737-900 | Installation of the Honeywell Enhanced Ground proximity Warning System (EGPWS) and Runway Awareness & Advisory System (RAAS). | | - Master Drawing 960-2227-002, revision CM, dated May 1, 2006, or later EASA approved revisions. | - Approval of this change in type design applies to the BOEING 737-200/ -300/ -400/ -500/ -700/ -800/ -900 aircraft equipped with either a flight management system (FMS) or a global positioning system (GPS) based navigation system or an EGPWS equippe (...) | 28.02.2007 |
| 10015934 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.02072 | | FAA STC ST00255DE-D | BAE 146/AVRO 146 SERIES | Installation of Allied Signal VHF AFIS | | FAA STC ST00255DE-D Learjet Inc Master Document List BAS46010001 Revision B dated May 30 1997 or later approved revision | -- | 26.09.2006 |
| 10015946 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.02088 | | FAA STC ST01000SE | BOEING 737-600 BOEING 737-700 BOEING 737-800 BOEING 737-900 | Installation of the Mark II Aircraft Communications Addressing and Reporting System (ACARS) in the Boeing 737-600, -700, -800, -900 Series Aircraft. | | - MDL 756-0124-001, Revision K, dated March 24, 2006, or later EASA approved revisions. | 1. Compatibility of this design change with other previously approved modifications must be determined by the installer. | 09.10.2006 |
| 10015949 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.02091 | | FAA STC ST01424CH-D | HAWKER 125-800 SERIES HAWKER 125-850XP SERIES | Installation of Honeywell N1 Digital Electronic Engine Control (DEEC) | | FAA STC ST01424CH-D Garrett Aviation Services Drawing List 43-80027-001 Revision C dated May 11 2006 or later approved revision. AFM Supplement 47-8027-001 Revision B dated May 11 2006 or later approved revision. For Hawker 800XP/850XP AFM Supplement (...) | | 04.10.2006 |
| 10015962 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.02110 | | FAA STC ST00844SE | BAE 146/AVRO 146 SERIES | Installation of the Flight Data Acquisition Management System (FDAMS) | | No change to existing documents | This change is not JAR-OPS-1.715 (EUROCAE ED55) compliant in the following parameters: Parameter 6, Pitch Attitude is only sampled once per second. Parameter 7, Roll Attitude is only sampled once per second. Parameter 23, Ground Spoiler position, Brake (...) | 13.12.2006 |
| 10016042 | HONEYWELL INTERNATIONAL Inc. | EASA.IM.A.S.02213 | | FAA STC ST01003CH-D | 600A, 600B, 700A, 700B DH 125 SERIES 1A, 3A, 3A/RA, F3B, F3B/RA, 400A, 400B, 403B, | Installation of Honeywell N1 Digital Electronic Engine Control (DEEC). | | Installation of Honeywell N1 Digital Electronic Engine Control (DEEC) in accordance with Garrett Aviation Services Drawing List DL1003, dated 26. Aug. 1999, no revision, or later approved revisions. Operation to be in accordance with an approved Airp (...) | 1. Compatibility of this modification with other previously approved modifications must be determined by the installer. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph (...) | 12.12.2006 |
| 10016089 | HONEYWELL INTERNATIONAL Inc. | EASA.IM.A.S.02276 | | FAA STC SA6086NM | BOEING 767-300 | Installation of Honeywell/Racal Aeronautical SATCOM | | Master Data List (MDL) ML4078149, Revision AE, dated December 19th, 2006 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 21.12.2006 |
| 10016115 | HONEYWELL INTERNATIONAL Inc. | EASA.IM.A.S.02306 | | FAA STC SA5473NM | BOEING 747-100-400 | Installation of Honeywell/Racal Aeronautical Satellite Communication system | | - Honeywell Master Data List, Engineering Specification No. ML4075075, Revision B, or later approved revisions of this specification, - Applicable Airplane Flight Manual Supplements are listed in Honeywell Master Data List, called out above. | Approval of this type design applies only to Boeing 747 series aircraft when: 1. Provisions for this SATCOM installation have been installed and approved in accordance with this STC, or 2. Provisions for this SATCOM installation have been install (...) | 10.04.2007 |
| 10016173 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.02400 | | FAA STC ST00717W1 | BOEING 737-200 SERIES BOEING 737-500 BOEING B737-300 BOEING B737-400 | Installation of the Honeywell Mark II ACARS | | Master Drawing List 159-08189-0001, Revision J, dated July 25, 2004, or later approved revisions Airplane Flight Manual Supplement 756-0182-900 revision "Original", dated September 16, 2004, or later approved revision. | Compatibility of this design change with previously approved modifications must be determined by the installer. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. | 21.05.2007 |
| 10016195 | HONEYWELL INTERNATIONAL Inc. | EASA.IM.A.S.02433 | | FAA STC ST00361LA | GULFSTREAM IV | Installation of the Honeywell HUD-2020 Head up Display System on Gulfstream G-IV aircraft for Category II approach criteria. | | In accordance with Master Data List Spec No 5220-0697, Revision G dated July 26, 1997 or later approved revisions. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 18.02.2009 |
| 10016402 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.02731 | | FAA STC ST00381SE | BOEING 757 | Installation of Enhanced Ground Proximity Warning System in B-757 Series Aircraft. Approval applies to aircraft equipped with a flight management system, which meets AC90-45A, or an EGPWS with a GPS based navigation system or an EGPWS equipped with an (...) | | Flight Manual Supplements approved by FAA 060-4189-002, 060-4189-015, 060-4189-029, 060-4189-022, 060-4189-055, 060-4189-054, 060-4189-068, 060-4189-065, 060-4189-081, 060-4189-082, 060-4189-079, 060-4189-117 Operating & Service Instructions* 060-4199-18 (...) | 1. Approval of this change in type design applies to the above model aircraft equipped with either a flight management system (FMS), which meets AC90-45A, or an EGPWS with a global positioning system (GPS) based navigation system or an EGPWS equip (...) | 11.04.2008 |
| 10028540 | HONEYWELL INTERNATIONAL INC. | EASA.IM.A.S.03210 | | | BOEING 747-400 BOEING 747-400F | | | | | 16.07.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|---------------------------|----------------|----------------------|--|---|--|---|--|---------------|
| 10031536 | HONEYWELL INTERNATIONAL INC. | P- EASA, I.M.A.S.01131 | | FAA STC ST01280SE | B737-600/-700/-800/-900 | Installation of the XK516 HF Transceiver System in accordance with Master Drawing List 756-0173-001 (and FAA STC ST01280SE dated 15 October 2003) | The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Master Drawing List ref. 756-0173-001, Revision F- Instructions for Continued Airworthiness Supplements as referenced in Master Drawing List ref. 756-0173-001 or later revisions of the above listed documents approved by EASA in accordance with EASA ED (...) | None | 24.08.2010 |
| 10034246 | HONEYWELL INTERNATIONAL INC. | | | FAA STC SR02114NY-D | EC135 P2(CPDS), EC135 P2+ EC135 T2(CPDS), EC135 T2+ | Installation of a Honeywell Model 15175 VXP System The Honeywell Model 15175 VXP System is a Health and Usage Monitoring System that monitors vibration characteristics of the aircraft to provide data to aid aircraft maintenance. The information provide (...) | The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Keystone Helicopter Installation Drawing List IDL01152, revision B- Honeywell Master Drawing List 159-08340-0001, revision -- Instructions for Continued Airworthiness: - Helicopter Instruction for Continued Airworthiness, Honeywell Chadwick Model (...) | None. | 28.03.2011 |
| 10038956 | HONEYWELL INTERNATIONAL INC. | | | FAA STC ST00806AT | B767-200/-300 | Installation of Aircraft Communications Addressing and Reporting System (ACARS) Mark II. | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this approval. | Master Drawing List 699-50097, revision M, 03/11/2005 AFM Supplement 756.0181.801 Revision 'A', 17/11/2005 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None. | 26.03.2012 |
| 10042984 | HONEYWELL INTERNATIONAL INC. | EASA, I.M.A.S.01874 | | FAA STC ST00440SE | A300B2K-3C, A300B4-2C A300B4-200, A300B4-600 A300F4-600 A310-200, A310-300 | Installation of Honeywell EGPWS on Airbus A300/A310/A300-600 aircraft models. This STC installs a Honeywell Enhanced Ground Proximity Warning System (EGPWS). | | Master Drawing List, ref: 960-2227-005, Revision AE, dated 11 October 2005 (or any later EASA approved revision). - Aircraft Flight Manual Supplement listed for the installation. - Instruction for Continued Airworthiness listed for the installation. | 1. Limitations on the STC EASA, I.M.A.S.01874 are the ones defined through the original FAA STC ST0440SE, at its latest amendment dated 27 May 2004 (or any later EASA approved revision). 2. This STC is approved only for the product configuration as defined (...) | 24.05.2007 |
| 10015981 | HORTON STOL-CRAFT INC. | EASA, I.M.A.S.02133 | | FAA STC SA989CE | CESSNA 170A CESSNA 170B | Installation of STOL Kit (Wing leading edge cuffs, drooped tips, stall fences and aileron gap seals) | | - In accordance with data and drawings called out by STOL-Craft Drawing List 1, or later approved revisions. - Operated in accordance with STOL-Craft Inc FAA approved Flight Manual Supplement dated January 29, 1974, or later approved revisions. | 1. Aircraft to be placarded "Intentional Spins Prohibited". 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine c (...) | 04.09.2006 |
| 10016763 | HOWARD J. FULLER, JR. | EASA, I.M.R.S.01092 | | FAA STC SR00081BO | SCHWEIZER 269 C-1 | Operation of rotorcraft on unleaded and leaded automotive gasoline, 91 minimum antiknock index (RON+MON)/2 per ASTM Specification D-439 or D-4814 as noted on placard to be installed in accordance with H.J.Fuller Jr. Installation Instructions No. HFJ (...) | | Howard J. Fuller Jr. Rotorcraft Flight Manual Supplement No. 269-1 Revision 2 dated July 2nd 2004 or later approved revisions. Installation Instructions No. HFJ97269, dated March 4, 1997 or later approved revisions. | 1. This STC is limited to those model helicopters equipped with Lycoming HO-360-C1A engines, modified in accordance with FAA STC No. SE2563CE.2. Howard J. Fuller, Jr. Rotorcraft Flight Manual Supplement No. 269-1, Rev 2 dated July 2, 2004 or later a (...) | 23.11.2005 |
| 10041028 | HOWARD J. FULLER, JR. | | REV. 1 | FAA STC SR00086BO | R44 | Operation of rotorcraft on unleaded automotive gasoline, 91 minimum antiknock index (RON + MON)/2 per ASTM Specification D-439 or D-4814 in accordance with Howard J. Fuller Installation Instructions No HJF97044. (This STC is based on FAA STCSR00086BO dated July 25, 2002) | The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Howard J. Fuller, Jr., Rotorcraft Flight Manual Supplement No. 44-1, Rev. 2, dated July 25, 2002 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) Howard J. Fuller Installation Instructions No HJF9 | 1. This approval is limited to those model helicopters equipped with Lycoming O-540-F1B5 engines modified IAW STC SE00653W (EASA, I.M.E.S.01002). This approval is applicable only with the following engine power ratings: - Maximum Continuous Power 205 HP (...) | 17.08.2012 |
| 10015608 | HYDRO-AIRE, INC | EASA, I.M.A.S.01544 | | FAA STC SA3339WE | BEECH 200, SUPER KING AIR KING AIR 200 | Installation of Power Brake / Skid Control System | | Model 200: Hydro-Aire Drawings No. 13-001 Revision C or later approved revision. Model B200: Hydro-Aire Drawings No. 13-001-100, Revision C or later approved revision. | 1. Models 200: Hydro-Aire Airplane Flight Manual Supplement dated on 17 MAY 1977 or later approved revision. 2. Models B200: Hydro-Aire Airplane Flight Manual Supplement No. R13-001-100-05 dated 12 MAY 2005 or later approved revision. 3. This STC is (...) | 02.09.2005 |
| 10014561 | IBERIA LINEAS AEREAS DE ESPANA SOCI | EASA, A.S.02661 | | | AIRBUS A321-212 | Passenger Cabin Lay-Out Modification for Special Flight | | IBE DOA0001 Rev 1, Dated 28th June 2006 | -- | 05.09.2006 |
| 10038270 | ICELANDAIR | | | | B757-200 | Installation of Satcom and Flight Tracking System. | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this approval. | Icelandair CRA-757-2320-43, Revision A, 30/01/2012 or later revisions of the above listed documents approved by EASA. | None. | 02.02.2012 |
| 10044500 | ICELANDAIR | | | | 757-200 | Communications - Data Transmission & Automatic Calling - DLink+ ACARS /CPDLC Installation Installation and activation of a Spectralux DLink+ system which provides Aircraft Communications Addressing and Reporting System (ACARS) /Controller Pilot Data LI (...) | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - Special Condition(s): CRI F-01 (Data Link Services for the Single European Sky) The requirements for environmental protection (...) | Change/Repair Approval, Icelandair document No. CRA 757-2320-52, Revision C (>,<-) dated 21 March 2013- Continued Airworthiness Instructions, Icelandair document No. CAI-757-2320-52 Revision B, 25 March 2013- Aircraft Flight Manual Supplement, Icela (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 17.04.2013 |
| 10044655 | ICELANDAIR | | | | B757-200 | Installation of AVIA Passenger Seats with IMS Rave IFE system and Astronics EMPOWER ISPS system (Installation of new seating layout with 183 total passengers using AVIO Interiors seats with IMS Rave IFE system and Astronics EMPOWER ISPS System Boeing 757-200 series of aircraft.) | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): CRI F-01 (Passenger Service, In-Flight Entertainment Systems and In-Seat Power Supply Systems) The req (...) | Change/Repair Approval, Icelandair document No. CRA-757-2520-381, Revision B (>,<-) dated 4 April, 2013.- Continued Airworthiness Instructions, Icelandair document No. CAI-757-2520-381->,<-) Initial Release, 1 April, 2013.- Aircraft Flight Manual S (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 26.04.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|----------------|----------------|-----------------------|-------------------------------|---|--|--|---|---------------|
| 10015254 | ICELANDAIR TECHNICAL SERVICES | EASA.A.S.03700 | | | BOEING 757-200 | Reconfigure seating layout to 52/10 Pax arrangement (VIP & Lounge) | | Change/Repair Approval, CRA-757-2520-276, Revision B, 28/11/2008Continued Airworthiness Instructions, CAI-757-2520-276,Revision IR, 30/09/2008 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.The i (...) | 15.12.2008 |
| 10029481 | ICELANDAIR TECHNICAL SERVICES | | | | B757-200 | Navigation - Dependant Position Determining Activation of the ADS-Bout function of the Transponder system. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.Applicant elected to comply with CS25 amt 7. | AFMS - AFM-757-3450-30 rev. AContinued Airworthiness Instructions CAI-757-3450-30 rev. BChange Approval CRA-757-3450-30rev. BCertification Plan 757-34-50-30 rev. BDrawing DW-G-757-3450-30-1 rev. BDrawing DW-G-757-3450-30-2 rev. Bor later revisions of the above listed documents approved by EASA | None. | 26.03.2010 |
| 10030954 | ICELANDAIR TECHNICAL SERVICES | | | | GA8 AIRVAN | Installation of TopScan Camera System | CS-23, Amdt. 1, is elected to comply with for the items affected by thechangeThe certificated noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval | EASA approved Master Document List, Doc. No. S21.MDL-0769, Issue 1,dated 07-Apr-2010, or later approved issueEASA approved Engineering Order, Doc. No. EO-GTA-2520-03, Rev. IR, dated28-Jun-2010, or later approved revision.EASA approved Flight Manual (...) | Only applicable for aircraft / preconditions as defined in EngineeringOrder, Doc. No. EO-GTA-2520-03, Rev. IR, dated 28-Jun-2010, or laterapproved revision. | 15.07.2010 |
| 10038368 | IMAGIK INTERNATIONAL CORPORATION | | | FAA STC NO. ST03400AT | BOEING 767-200/-300 | Installation of IMAGIK LCD Entertainment Monitors. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | FAA STC No. ST03400AT dated 25 February 2008;Avionics Support Group Master Data List Drawing 26012-MDL, Revision Bdated15 May 2007;or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions ofthis decision). | None. | 09.02.2012 |
| 10015207 | INAER AVIATION ITALIA S.p.A. | EASA.A.S.03620 | REV. 1 | | CESSNA 500 | STC EASA.A.S.03620 dated 2 June 2009 "Satellite Telephone System" | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS-25. 1309 | Master Drawing List MDL-23-00-0802-A dated 26 May 2009 | None | 23.02.2012 |
| 10032602 | INAER AVIATION ITALIA S.p.A. | | REV. 1 | | CESSNA 650 (CITATION VI) | Enhanced Ground Proximity Warning System (EGPWS) Honeywell MK VIIIInstallation.Honeywell EGPWS MK VIII Installation in replacement of Honeywell GPWS MKVI. | FAA TCDS A9NMThe Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunch (...) | -Master Drawing List MDL-34-44-1003-A, Revision 00 from 10 Nov. 2010-Aircraft Flight Manual Supplement FMS-34-44-1003-A, Revision 00 from 18May 2010-Instructions for Continued Airworthiness ICA-34-44-1003-A, Revision 00from 04 June 2010-SAN statemen (...) | STC is applicable to S/N 650-0216 only | 26.09.2011 |
| 10038266 | INAER AVIATION ITALIA S.p.A. | | | | CL-215-6B11 (CL415 VARIANT) | Elementary Surveillance Mode S Diversity TranspondersUpgrade/Replacement and Installation of new transponders with Elementary Surveillance and Diversity capabilities. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | INAER Master Drawing List Doc. No. MDL-34-59-0912-A Rev. 01dated 19January 2012;INAER Aircraft Flight Manual Supplement Doc. No. FMS-34-59-0912-A Rev. 0dated 09 January 2012;INAER Instruction for Continued Airworthiness ICA-34-59-0912-A Rev. 0dated (...) | None. | 01.02.2012 |
| 10044217 | INAER AVIATION ITALIA S.p.A. | | | | AB 412 EP | EMS Configuration as per Master Drawing List MDL-25-00-1213-A, revision00, dated 22 March 2013. | The Certification Basis (CB) for the original product remains applicabto this certificate/ approval.The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement FMS-25-00-1213-A, revision 00, dated07 March 2013;or later revisions of the above listed documents approved by EASA.Instruction for Continued Airworthiness ICA-25-00-1213-A, revision 01,dated 13 March 2013. | Prior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noadverse effect upon the airworthiness ofthe product. | 26.03.2013 |
| 10017095 | INAER MAINTENANCE S.A.U. | EASA.R.S.01010 | | | 212, 412, 412EP, AB212, AB412 | Installation of external Cargo basket for "Bambi Bucket" | | Airworthiness Document CMA005.AAA E1 R0Design Report CMA005-ID E1 R0 (Flight Test)Production Document CMA005-PROD E1 R0Maintenance/Installation Instructions Document CMA005-MID E1R0RFM Supplement CMA005AFMS E1 R0 | In accordance with RFM CMA005AFMS E1 R0 Supplement | 23.05.2006 |
| 10030130 | INAER MAINTENANCE S.A.U. | | | | BELL 412EP | Installation of Weather Radar Bendix-King model RDR1400C inBell 412EPHelicopters. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | INAER 210-LMD Rev. 0 (Master Document List)INAER 210-BI, Rev. 0 (Engineering Bulletin)INAER 210-FMS, Rev. 0 (Supplemento RFM)INAER 210-ICA, Rev. 0 (Instructions for Continued Airworthiness)or later revisions of the above listed documentsapproved by EASA | None. | 26.05.2010 |
| 10034643 | INAER MAINTENANCE S.A.U. | EASA.R.S.01019 | | | AS 365N3 | Adaption of a FLIR ULTRA 4000 system installation provisions for a FLIRULTRA FORCE II. This modification is a minor change of the STC FLIRULTRA 4000 - 365N3 REF. AHSE-50300 of ASTEC HELICOPTER SERVICES AS,keeping the supporting structure and the electrical provisions for FLIRULTRA 4000. | | Airworthiness Document CMA034-AAA E1 R0Engineering BulletinCMA034-BI E1 R0Maintenance and Installation/ UninstallationInstructions CMA034-MID E1R0Drawings ZW-4003-01 A HELICSA, ZW-2010-01 H, ZW-2009-01 H2 & VCR FOR LEOIIAFM Supplement CMA034-AFMS E1 (...) | FLIR ULTRA 4000 must be previously fitted according to the STC FUR ULTRA4000 - 365N3 REF. AHSE-50300 of ASTEC HELICOPTER SERVICES AS | 20.05.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|---------------------|----------------|------------------------|--|--|---|--|---|---------------|
| 10035000 | INAER MAINTENANCE S.A.U. | | | | SA330J | Modification of transponder wiring installation to comply with EASA ADNo 2010-0067 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master document list INAER.M11/010/34-LMD, Revision 0-Engineering bulletin INAER.M11/010/34-BI 0, Revision 0-Declaration of compliance INAER.M11/010/34/DoC-1 dated 17 May 2011 | None | 18.05.2011 |
| 10040333 | INAER MAINTENANCE S.A.U. | | | | P.68 C P.68 OBSERVER P.68 OBSERVER 2 | "External load mount". Installation of a mount to hold external loads in the existing photogrammetric hatch, according to Certification Plan "INAER.M11/037/25-PdC", Revision 0, Date: 21/06/2012. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Technical Documents to be Approved by EASA: 1., Master Document List INAER.M11/037/25-LMD, Revision 0, Date: 21/06/2012. 2., Instructions for continued airworthiness INAER.M11/037/25-ICA, Revision 0, Date: 21/06/2012. 3., Flight manual supplement INAER.M (...) | none | 26.06.2012 |
| 10040926 | INAER MAINTENANCE S.A.U. | | REV. 1 | | B200GT | "Stretcher kit Avfab 32-0117". Installation of a stretcher kit replacing LH seats (rows 2 and 3), according to Certification Plan "INAER.M12/044/25-PdC", Revision 0, Date: 27/07/2012. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 1. Master Document List INAER.M12/044/25-LMD, Revision 0, Date: 27/07/2012. 2. Instructions for continued airworthiness INAER.M12/044/25-ICA, Revision 0, Date: 27/07/2012. 3. Flight manual supplement INAER.M12/044/25-FMS, Revision 0, Date: 27/07/2012. or (...) | None | 08.08.2012 |
| 10043924 | INAER MAINTENANCE S.A.U. | | | | MBB BK117 B-2 | Adaptation of FLJR operator desk and aft seat. In order to operate the FLJR with the long range fuel tanks (TC holder optional installation no 801), the aft seats have to be moved forward and modified to fit in the new attachment points. Also the opera (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement CMA041 - FMS R0 or later revisions of the above listed documents approved by EASA Master document list CMA041 - LMD R0 Engineering bulletin CMA041 - BI R0 Instructions for Continuous Airworthiness CMA041 - ICA R0 | Only applicable to s/n 7176 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the (...) | 05.03.2013 |
| 10029799 | INDIGOSAT | | | | 208 / 208B | Installation of IndigoOne Satellite Tracking and Communication System (Kit Part No. IGO1_MA1236_01R1-1) in accordance with CASA (Australia) STC SVA 513 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Manual IS-IM-2007-001, Issue B, dated 27. February 2008 or later revisions of the above listed documents approved by EASA | Effectuated S/N see in Installation Manual IS-IM-2007-001, Issue C | 21.04.2010 |
| 10016562 | INFINION CERTIFICATION ENGINEERING | EASA.IM.A.S.02948 | REV. 1 | TCCA STC SA06-30 | BAE JETSTREAM 3100 | Concorde RG-380 Battery Installation. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | # Master Data List: MDL-1090-01;# Installation Instructions EO-1090-01;# Flight Manual Supplement FMS01090-01;# Technical Note: Substitution of Battery Replacement. TN-1063-01;# Compliance Program: CP-1090-01;# Maintenance Manual Supplement MMS-1090-01. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 16.09.2008 |
| 10046122 | INFINION CERTIFICATION ENGINEERING | | | TCCA SA06-59 | PA-31-350 (CHIEFTAIN) | Fuselage Camera Hole Cutout - Installation of fuselage camera hole, including 20 inch x 20 inch opening and rerouting of elevator trim and rudder trim cables. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Data List No. MDL-1141-01, issue 4, dated 12 December 2010; - PA-31-350 Flight Manual Supplement No. FMS-1141-01, issue 1, dated 12 September 2006; - PA-31-350 Maintenance Manual Supplement No. MMS-1141-01, issue 2 (>-<) 7 July 2009 or later (...) | EASA has not certified the equipment for which these provisions are intended. You must get an additional EASA Approval to install such equipment. The installer must determine whether this design change is compatible with previously approved modifications (...) | 15.08.2013 |
| 10015538 | INFLIGHT CANADA Inc. | EASA.IM.A.S.01422 | REV. 1 | TCCA STC C-LSA95-037/D | AIRBUS A320-232 | SONY TRANS COM Passenger Entertainment System installation. Note: This STC had to be revised due to a correction to the holder of the original STC. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Modification Data Summary ref. A71200 revision G or later EASA approved revision | 1. This STC is limited to the A320 MSN's 496 and 5252. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 27.11.2006 |
| 10016609 | INFLIGHT CANADA Inc. | EASA.IM.A.S.03029 | | TCCA STC O-LSA03-127/D | BOMBARDIER CL-600-2B19 | Honeywell SCS-1000 Mini-Aero SATCOM System and Securaplane Camera Installation. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Transport Canada SoC ref. 5010-O-08-0561 dated 22 October 2008. - Transport Canada STC No. O-LSA-03-127/D Issue 3 dated 16 October 2004. | 1. DECA Aviation / Inflight Canada Modification Summary MS03055, Revision 4, dated 16 October 2004, or later approved revision. 2. DECA Aviation / Inflight Canada Instructions for Continued Airworthiness ICA03055, Revision 2, dated 12 October 20 (...) | 04.12.2008 |
| 10027294 | INFLIGHT CANADA Inc. | P-EASA.IM.A.S.03187 | REV. 4 | TCCA STC SA09-28 | 737-800 | Rockwell Collins dPAVES Overhead Passenger Entertainment System, TCCASTC Number: SA09-28 Issue 3 dated Nov 16, 2009. This revision consists in removing the limitation on EWIS ICA | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Certification Basis in accordance with the EASA TCDS IM.A.120 plus: CRIH01/Special Con (...) | Inflight Canada / DECA Aviation Documents: Master Data List MDL08119-STC Rev.12 Airplane Flight Manual Supplement FMS08119-STC Rev.2 Instructions for Continued Airworthiness ICA08119-STC Rev.3 or later revisions of the above listed documents approved by (...) | none | 04.06.2012 |
| 10030534 | INFLIGHT CANADA Inc. | | REV. 1 | TCCA STC O-LSA08-045/D | 767-300 | Audio Video on Demand (AVOD) and In-Seat Power Supply (ISPS) System Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. As per the EASA TCDS including EWIS ICA requirements (CRIH-01) The requirements for environmental (...) | Master Data List, MDL05217, Revision 11, 29/03/2010 Airplane Flight Manual Supplement, FMS05217, Revision 3, 25/02/2010 EWIS ICA, EWIS ICA05217, Revision 1, 14/11/2011 or later revisions of the above listed documents approved by EASA in accordance with (...) | none | 10.09.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|-------------------------------|--|--|---|---|--|---------------|
| 10041380 | INFLIGHT WARNING SYSTEMS, Inc. | | | FAA STC ST02300LA | B777-200, 200LR, 300, 300ER | Validation of FAA STC ST02300LA related to Installation of Fan Vibration/Monitor Unit. | The Certification Basis for the original product remains applicable to this certificate/ approval. | Master Drawing List 217700, Revision D, dated May 21, 2010; Instruction for Continued Airworthiness 13705, Revision A, dated May 21, 2010; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None. | 12.09.2012 |
| 10041522 | INFLIGHT WARNING SYSTEMS, Inc. | | | | A318-111/-112/-121/-122 A319-111/-112/-113/-114/-115 A319-131/-132/-133 A320-111/-211/-212/-214 A320-231/-232/-233 A321-111/-112/-131/-211/-212 A321-213/-231/-232 | Installation of Inflight Warning System Fan Vibration Monitor Unit (VMU) model n. 216001400 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -FAA Supplemental Type Certificate ST02356LA issued 5th December 2011- PSCP-216001 Rev. G, dated 17th May 2011- Master Data List 13601 Rev. A, dated 31st May 2011- Instructions for Continued Airworthiness 13608 Rev. A, dated 25th April 2011 or later rev. (...) | None | 25.09.2012 |
| 10017143 | ING. MIROSLAV RUSINAK S.R.O. | EASA.R.S.01246 | | | BELL 427 | Installation of the articulated patient loading system into Bell 427 rotorcraft. | | 06042101r/RMS dated 8 February 2007 # flight manual supplement 06042401r/MDS dated 31 January 2007 # Manufacturers Data Supplement 06052301r/MM dated 21 April 2006 # Maintenance Manual | 427-706-020-101 # Bell Helicopter Textron Cabin Hardpoints Provision Kit must be installed (BHT-427-II-27 rev 0 dated 16 September 2004) prior to installation of this STC. STC EASA.IM.R.S.00548 or EASA.IM.R.S.00577 (FAA STC SR02381A) # Installation (...) | 19.02.2007 |
| 10038518 | ING. MIROSLAV RUSINAK S.R.O. | P-EASA.R.S.01416 | | | BELL 427 | Support unit for medical devices | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 07100103r initial release - Flight Manual Supplement or later revisions of the above listed documents approved by EASA 07100101r - Compliance Checklist 07100102r initial release - Maintenance Manual 07100108r initial release - Mechanical Work Order (con (...)) | None | 22.02.2012 |
| 10027603 | INGENIEURBUERO FUER LUFTFAHRT | P-EASA.R.S.01447 | | | AS350 B1, B2, B3 AS355 E, F, F1, F2, N, NP | Installation of airborne laser scanning equipment in AS350/355 series helicopter | FAR 27, Amendment 21 The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | -Engineering Summary Report: ESR-35X-0801, Rev. 0.4-Rotorcraft Flight Manual Supplement: "Installation of Airborne Laser Scanner System" FMS-35X-0801, Rev. 5-Installation Instruction: IIS-35X-0801, Rev. 2-Instructions for continued airworthiness: ICA- (...) | Service Bulletin No. 25.00.62 is a prerequisite and must be installed prior to this modification. Any alteration of installed equipment leads to non-compliance of committed and approved design and therefore requires reinvestigation and change approval prior use. | 13.10.2009 |
| 10015589 | INNTECH AVIATION | EASA.IM.A.S.01500 | | TCCA/LSTC Q-LSA05-029/D | BOMBARDIER BD-700-1A10 | Miscellaneous modifications to cabin interior and installation of Electronic Flight Bag System provisions for BD700-1A10 Global Express Acc.: MDL-05103002 Rev. B or later TCCA approved rev. Validation of imported TCCA/LSTC: Q-LSA05-029/D | | FCOM Supplement: FOMS05103005 Rev A | 1. Limited to S/N 90172. Regarding the installation of the Electronic Flight Bag System the design covers only the mounting device and its connection to the aircraft. The suitability, integrity or accuracy of the EFB hardware, software, databa (...) | 18.11.2005 |
| 10015904 | INNTECH AVIATION | EASA.IM.A.S.02030 | | TCCA LSTC Q-LSA06-026/D ISS 1 | BD-700-1A10 | Modification of EGPWS Radar Altimeter Callouts and installation of the IPOD on ACSN 9163/Validation of TCCA Limited STC Q-LSA06-26/D to S/N 9163MDL: 26126002 Rev NC or later approved revisions TCCA SoC: NAI5012-06-0063 dated June 20, 2006 | | | | 28.07.2006 |
| 10016130 | INNTECH AVIATION | EASA.IM.A.S.02324 | | TCCA STC Q-LSA06-018D | BOMBARDIER BD-700-1A10 | Re-Locate Electronic Flight Bag and Miscellaneous Cabin Modifications in Cockpit of SN 9145/Validation of TCCA Limited STC Q-LSA06-018/D to S/N 9145MDL: 26115002 Rev A or later TCCA approved revisions Compliance Plan: CP26115-001 | | IFCA: TL26115-004 Rev NC or later TCCA approved revisions | | 22.03.2007 |
| 10016228 | INNTECH AVIATION | EASA.IM.A.S.02480 | | TCCA STC Q-LSA07-029/D | BD-700-1A11 | Installation of Photosmart printer and installation of additional handset for in the aft fhs of SN 9204/Validation of TCCA Limited STC Q-LSA07-029/D to S/N 9204 | | Compliance plan: CP27116-001. | | 21.01.2008 |
| 10016277 | INNTECH AVIATION | EASA.IM.A.S.02566 | | Q-LSA07-035/D | BD-700-1A11 | Miscellaneous Cabin Modifications on SN 9189/Validation of TCCA Limited STC Q-LSA07-035/D to S/N 9189MDL: MDL27124002 Compliance Plan: CP27124-001 Rev. A | | IFCA: TL27124-004 Soc: NIA5012-07-0128, dated August 28, 2006 | | 12.10.2007 |
| 10016407 | INNTECH AVIATION | EASA.IM.A.S.02737 | | TCCA STC Q-LSA08-001/D | BOMBARDIER BD-700-1A10 | Replacement of conference table in aircraft S/N 9197, aft cabin. | | Transport Canada LSTC No. LSA08-001/D Issue 1 dated 15 January 2008. Transport Canada Statement of Compliance ref: NAI5012-08-0003 dated 18 January 2008 MDLZ7142002, Master Document List - Conference Table Replacement on Aircraft S/N 9197, revision NC, (...) | Restricted to ASN 9197. With this modification incorporated Aft LSH forward facing double seats must not be occupied during taxi take off and landing. | 19.03.2008 |
| 10016508 | INNTECH AVIATION | EASA.IM.A.S.02875 | | TCCA Q-LSA08-021/D | BOMBARDIER CL-600-2B16 (CL-604) | Cabin Finishing Refurbish on aircraft S/N 5442. | | 1. TCCA STC No. Q-LSA08-021/D Issue 3 dated 11 June 2008. 2. Innotech Aviation Master Document List MDL28103002 revision B dated 11 June 2008. | 1. Restricted to ASN 5442.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airwor (...) | 06.10.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------|--------------|----------------|----------------------------|------------------------------|--|--|--|---|---------------|
| 10031405 | INNTECH AVIATION | | REV. 1 | TCCA STC O-LSA10-185/D | BD-700-1A10, ASN 9287 HB-JGE | H-Ku System Installation with Viasat VMT 1500 Ku Band System and Honeywell MHC Cabin Management. | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | Transport Canada STC No. O-LSA10-185/D Issue 1 dated 23 June 2010. Installation must be in accordance with Innotech Aviation Drawing List MDL-9287-156 Revision F, or later approved revision. Operation must be in accordance with Innotech Aviation Airpla (...) | Restricted to ASN 9287 | 23.05.2011 |
| 10031686 | INNTECH AVIATION | | | TCCA STC SA10-18 | BD-700-1A10 | I-CAIR System Installation with JetAir Bio-Protection (BPS) System | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | Transport Canada STC No. SA10-18 Issue 1 dated 09 February 2010. Transport Canada SoC ref. 5010-O-10-0427 dated 31 May 2010. Innotech Aviation Drawing List MDL-BD700-151 Revision NC dated 15 January 2010. Innotech Aviation Airplane Flight Manual Supplement (...) | none | 03.09.2010 |
| 10031791 | INNTECH AVIATION | | REV. 2 | TCCA STC N° C-LSA10-022/D | BD-700-1A10 | Installation, Complete Custom Aircraft Interior | EASA TCDS No. IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product (...) | EASA STC No. 10031791. TCCA SoC ref. C-10-0936 dated 14 December 2010. TCCA STC No. C-LSA10-022/D Issue 1 dated 06 August 2010. ACS-NAI Ltd. Modification Data Summary G481000 Revision B. Flight Manual Supplement for Interior Completion, FMS-IA700-9352Rev (...) | Restricted to ASN 9352.1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2. WLAN system is compatible only to IEEE 802.11(b). | 31.01.2011 |
| 10032531 | INNTECH AVIATION | | REV.1 | TCCA STC NO. C-LSA10-207/D | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9374 | EASA TCDS No. IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. Non-significant. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the o (...) | TCCA SoC ref. C-10-1069 dated 12 January 2011. TCCA STC No. C-LSA10-207/D Issue 1 dated 05 November 2010. ACS-NAI Ltd. Modification Data Summary G902000 Revision NA. Flight Manual Supplement for Interior Completion, FMS-IA700-9374 Revision NC. Flight Man (...) | Restricted to ASN 9374.1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2. WLAN system is compatible only to IEEE 802.11(b). | 17.01.2011 |
| 10035813 | INNTECH AVIATION | | | TCCA STC Q-LSA06-059/D | BD-700-1A10 | Cabin Management System Upgrade - ASN 9020 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | Transport Canada STC No. Q-LSA06-059/D Issue 4 dated 31 August 2009. Kmlcx Aviation Master Document List No. MDL26147002 - Cabin Management System Upgrade on Global Express, ASN 9020, Revision C, dated 31 August 2009 or later approved revision. Kmlcx (...) | Restricted to ASN 9020 | 19.07.2011 |
| 10036090 | INNTECH AVIATION | | | TCCA STC O-LSA07-336/D | BD-700-1A10 | ICG BD700 Satcom Upgrade. | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | TCCA STC No. O-LSA07-336/D Issue 1 dated 18 December 2007. Innotech Aviation, a Division of IMP Group Ltd. Master Drawing List MLGX-9028-126 Revision N/C, dated 18 December 2007, or later approved revision. Maintenance must be in accordance with Innotech (...) | Restricted to ASN 9028. | 10.08.2011 |
| 10036091 | INNTECH AVIATION | | | TCCA STC O-LSA08-197/D | BD-700-1A10 | Software and DLNA Upgrade to Existing HSB 400 Satcom Installation. | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | TCCA STC No. O-LSA08-197/D Issue 1 dated 29 August 2008. Avionics Design Services Ltd. Master Drawing List MDL08080 Revision N/C dated 29 August 2008 or later approved revision or later revisions of the above listed documents approved by EASA in accordance (...) | Restricted to ASN 9028. | 10.08.2011 |
| 10036347 | INNTECH AVIATION | | | TCCA STC C-LSA11-079/D | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9399 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | TCCA STC No. C-LSA11-079/D Issue 1 dated 24 May 2011. ACS-NAI Ltd. Modification Data Summary H113000 Revision NC or later approved revision. Innotech Aviation Flight Manual Supplement for Interior Completion, FMS-IA700-9399 Revision NC or later appro (...) | Restricted to ASN 9399. The wireless Local Area Network (WLAN) System is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDS). Any other intended function of this equipment will require a re-examination of the certification basis. | 02.09.2011 |
| 10036823 | INNTECH AVIATION | | | TCCA STC NO. SA10-8 | BD-700-1A10 | H-Ku System Installation with VIASAT VMT 1500 KU Band System | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | Transport Canada STC No. SA10-8 Issue 1 dated 14 January 2010. Installation must be in accordance with Innotech Aviation Drawing List MDL-BD700-148 Revision NC, dated 01 December 2009 or later approved revision. Operation must be in accordance with Innote (...) | None | 10.10.2011 |
| 10037477 | INNTECH AVIATION | | | | BD-700-1A10 | Installation, Complete Custom Aircraft Interior - ASN 9416 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | TCCA STC No. C-LSA11-120/D Issue 1, dated 13 October 2011. ACS-NAI Ltd. Modification Data Summary H164000 Revision NC or later approved revision. Flight Manual Supplement for Interior Completion, FMS-IA700-9416 Revision NC or later approved revision. Fligh (...) | Restricted to ASN 9416. Refer to the Airworthiness Limitations contained in the required Airplane Flight Manual Supplements. The Wireless LAN System is intended to provide internet connectivity and email services to the airplane's cabin using portable e (...) | 29.11.2011 |
| 10040515 | INNTECH AVIATION | | | TCCA STC NO. Q-LSA07-040/D | BD-700-1A10 | Installation of Iridium Phone System and interior modification - ASN9028 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | Transport Canada STC No. Q-LSA07-040/D Issue 3, dated 07 June 2011. Innotech Aviation MDL27127002 Master Document List - Installation of Iridium Phone System and Minor Interior Modifications on ACSN 9028, Revision B, dated 07 June 2011. Innotech Aviation (...) | Restricted to ASN 9028. Previous installation of EASA STC No. 10039790 and EASA STC No. 10039791 are pre-requisites. | 06.07.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|---------------------|----------------|----------------------|--|---|---|---|---|---------------|
| 10040920 | INNTECH AVIATION | | REV. 1 | TCCA STC C-LSA12-105 | BD-700-A110 | Installation, Complete Custom Aircraft Interior - ASN 9433 | EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged (...) | Transport Canada STC No. C-LSA12-105 Issue 1 dated 09 July 2012. ACS-NAI Ltd. Modification Data Summary H430000 Revision NC. Innotech Aviation Flight Manual Supplement for Interior Completion, AFMS-IA6000-9433 Revision BASIC. For Instructions for Continu (...) | Restricted to ASN 9433. The Wireless LAN System is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require re-examination of the certification basis. | 08.08.2012 |
| 10043232 | INNTECH AVIATION | | | TCCA STC SA12-63 | BD-700-1A10 | Installation, Aircraft Completion Modifications. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS25 Amendment 1 - Various. The requirements for environmental protection and the associated certified noise and/ or emissions (...) | Transport Canada STC No. SA12-63, Issue 1, dated 17 October 2012. Innotech Aviation Configuration Definition Record CER-BD700-001 Revision A. Innotech Aviation Airplane Flight Manual Supplement AFMS-IA6000-2500 Revision A. Instructions for Continued Airwor (...) | Restricted to ASN 9482 - The WLAN System is intended to provide internet connectivity and e-mail services to the airplane's cabin using Portable Electronic Devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis. | 14.01.2013 |
| 10046109 | INNTECH AVIATION | | | TCCA STC SA12-63 | BD-700-1A10 BD-700-1A11 | Installation, Aircraft Completion Modifications. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - Special Condition(s): CRI D-02 Issue 2, Side Facing Divans - Equivalent Safety Finding(s): CRI D-04 Issue 2, Emergency E (...) | Transport Canada STC No. SA12-63 Issue 1, dated 17 October 2012. Innotech Aviation Configuration Definition Record CER-BD700-001 Revision E. Innotech Aviation Airplane Flight Manual Supplement AFMS-IA6000-2500 Revision A. Innotech Aviation Instructions f (...) | EASA approved configurations are identified in Configuration Definition Record CER-BD700-001. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (...) | 14.08.2013 |
| 10033170 | INNOVATIVE SOLUTIONS AND SUPPORT, I | P-EASA.IM.A.S.03020 | | FAA STC ST0237ZCH | B757-200/-300 | Installation of Integrated Flat Panel Display System (IFPDS) and L-3 Avionics Systems GH-3100 Electronic Standby Instrument System (ESIS) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate. For Changed Areas the Certification Basis is CS 25 Initial Issue. Applicable Requirements are: 25.3 (...) | Master Data List, 1D-07687, Revision 11, December 9th, 2010. Instructions for Continued Airworthiness, 1D-07688, revision 8, June 7th, 2010. Aeroplane Flight Manual Supplement, 1D-07718, Revision 5, July 28th, 2010 or later revisions of the above listed documents (...) | Approved Installation Configuration 9 only | 21.12.2010 |
| 10036223 | INNOVATIVE SOLUTIONS AND SUPPORT, I | P-EASA.IM.A.S.03107 | | FAA STC ST02165NY | B767-200/-300/-300F | Installation of Flat Panel Display System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate. For changed areas the certification basis is CS25 Initial Issue; 25.301 - 25.303 - 25.305 - 25.30 (...) | Master Data List (MDL), Document 1D-07792, rev 11, 15/03/2011 («-») Instructions for Continued Airworthiness (ICA), Document 1D-07793, Revision 6, 14/01/2011 Aeroplane Flight Manual Supplement (AFMS), Document 1D-07125, Revision 5, 28/07/2010 or later re (...) | Installation Configuration 6 approved only. The Smiths Aerospace Limited Integrated Standby Instrument System (ISIS) is not approved for installation under this STC. No earlier revisions of the listed documents are approved under this STC. | 23.08.2011 |
| 10043056 | INTEC HELICOPTERS Inc. | EASA.IM.R.S.0030 | | | Data not found | Installation of Engine air filter in accordance with INTEC Master Drawing List No. 1206 No. 1206-1001 Revision F1, dated 30 JAN 2004 (or later EASA Approved revision). | | | 1. A copy of this STC and the appropriate approved Flight Manual Supplement (see INTEC Master Drawing List N° 1206-1001) must be maintained as part of the permanent records for the modified aircraft. Additionally, for Bell Model 206-A, B and B1 aircraft (...) | 09.07.2004 |
| 10016867 | INTEGRATED FLIGHT SYSTEMS, INC | EASA.IM.R.S.01243 | | FAA STC SH3509SW | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 C EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 D1 EUROCOPTER FRANCE EC 130 B4 | Installation of a Belt Driven Freon Air Conditioning System (FAA STC SH3509SW). | | 1. VFR Operations only. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/eng (...) | 26.04.2007 | |
| 10029611 | INTEGRATED FLIGHT SYSTEMS, INC | P-EASA.IM.R.S.01407 | REV. 1 | STC SR09178RC | 600N | Integrated Flight Systems Vapor Cycle Air Conditioning System | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Drawing list DL-44 Rev B, 15th March 2001. Electrical routing Drawing 2-MDHS-600N-01 Structural Substantiation report FS-600S-02 Structural Substantiation report IFS-600S-02 Structural Substantiation report IFS-600S-03 Structural Substantiation report (...) | None | 29.06.2010 |
| 10040183 | INTERNATIONAL ENGINEERING ASSISTANC | | | | F177RG | IEA/STC/ERHC/01/2010 revision 00 - Electric Rudder Hand Control. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 23 The certified noise and/ or emissions levels of the original product are unchanged (...) | Supplément au manuel de maintenance SUP-ERHC-MANUEL DEMAINTEENANCE-CESSNA177RG-2010 issue D; Supplément au manuel de vol SUP-ERHC-MANUEL DE VOL-CESSNA177RG Issue G; or later revisions of the above listed documents approved by EASA. | Limitation to F177RG serial number 0016. | 15.06.2012 |
| 10027431 | ISEI | P-EASA.R.S.01543 | | | AS350 B3, EC130 B4 | -Installation of Safetyplane Equipment- Installation of a system for tracking a set of flight and system parameters during operation of the rotorcraft. The modification is based on ISEI change N-EC130 AS350B3 as defined in Dossier de Définition de la Modification-F-A-03, dated 26.06.2009. | EASA Certification Basis as per EASA TCDS.R.008 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The change complies with CS27 First Issue T (...) | Dossier d'approbation de modification N-EC130 AS350B3 Approbation-F-A-02, dated 23.07.2009-Dossier de Définition de la Modification N-EC130 AS350B3 Définition-F-A-03, dated 26.06.2009-Dossier d'Installation N-EC130 AS350B3 Installation F-A-04, dated 2 (...) | None | 30.09.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------|------------------|----------------|----------------------|--|---|--|--|--|---------------|
| 10027889 | ISEI | | REV. 1 | | EC 120 B | Installation of HELICOM V1 or HELICOM V2 data recording system. Revision 1 of STC 10027889 introducing change in system name (Safetyplane becomes HELICOM V1) <-> new V2 version and associated documentation. | The Certification Basis for the original product remains applicable to this certificate/ approval. The Certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Design File n° N3-EC120-HV2 NUM E-A-03, dated 18 May 2011. Installation file n° N7-EC120-HV2 NUM E-A-03, dated 18 May 2011. Installation file n° N4-EC120 - Installation E-A-07, dated 21 October 2009. Maintenance Manual Supplement n° N8-EC120-HV2 NUM E-A-0 (...) | None. | 03.06.2011 |
| 10028200 | ISEI | | | | R44, R44 II | Title: Installation of Safetyplane Equipment Description: Installation of a system for tracking a set of flight and system parameters during operation of the rotorcraft. The modifications based on ISEI change N-R44 as defined in the Master Document L (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Instructions, N6-R44-Installation Document E-A-02- Instructions for Continued Airworthiness, N7-R44-Maintenance manuals supplement E-A-02- N2-R44-Approval Change Document E-A-02- N3-R44-Experience Feedback E-A-02- N4-R44-Safety Assessment (...) | None | 08.12.2009 |
| 10028789 | ISEI | | | | AS350 B, BA, BB, B1, B2 | Installation of Safetyplane Equipment Installation of a system for tracking a set of flight and system parameters during operation of the rotorcraft as defined in the Approval Change Document N2-AS350 B BA BB B1 B2 E-A-02. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The change complies with CS27 First Issue. The certified noise and/or emissions level (...) | Approval Change Document N2-AS350 B BA BB B1 B2-Approval changed document, E A 02, 02/02/10 or later revisions of the above listed documents approved by EASA-Safety Assessment Document N4-AS350 B BA BB B1 B2-Safety assessment document, E A 02, 02/02/10 (...) | None | 09.02.2010 |
| 10033288 | ISEI | P-EASA.R.S.01542 | REV. 3 | | AS 355 NP | INSTALLATION OF HELICOM V2- NUM Installation of a System for Tracking and Recording a set of flight and system parameters during operation of the rotorcraft Installation of a System for Tracking and Recording a set of flight and system parameters during o (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval (*) AS355NP CRI F-01, Special Condition for Lithium Battery Installation, dated 17/05/20 (...) | Design change N3-AS355 NP-HV2 NUM-E-B-03- Installation Manual N7-AS355 NP-HV2 NUM-E-B-03- Maintenance Manual Supplement N8-AS355 NP-HV2 NUM-E-B-03 | The system does not replace standard maintenance records and pilot reporting. Any broadcasting must be done in accordance with National regulations. The approval holder shall fulfill the obligations of Part 21-> <-> Paragraph 21A.109. Prior to installation (...) | 07.10.2013 |
| 10033974 | ISEI | | REV. 1 | | A109E | Installation of HELICOM V1 or HELICOM V2 data monitoring system. Revision 1 of STC 10033974 introducing change in system name (Safetyplane becomes HELICOM V1), new V2 version and associated documentation. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Maintenance Manual Supplement N8-A109E-HV1 E-A-06, dated 16 February 2011. Installation File N7-A109E-HV1 E-A-06 dated 16 February 2011. Maintenance Manual Supplement N8-A 1 09E-HV2 NUM E-A-02, dated 24 March 2011. Installation File n° N7-A109E-HV2 NUM E-A-02, dated 24 March 2011 | None | 05.05.2011 |
| 10034914 | ISEI | P-EASA.R.S.01544 | REV. 1 | | EC135 P1 (CPDS) EC135 P2 (CPDS) EC135 P2+ EC135 T1 (CPDS) EC135 T2 (CPDS) EC135 T2+ | Installation of HELICOM V2 NUM Installation of a system for tracking a set of flight and system parameters during operation of the rotorcraft. The modification is based on ISEI change "Installation of HELICOM V2 NUM" as defined in Approval Change File N1- (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 27 Initial Issue for the change. EASA Special Condition F-01 on Lithium Battery Inst (...) | Approval Change File N1- EC135 V2-HV2 NUM E-A-07- Installation Instructions N7- EC135 V2-HV2 NUM E-A-07- Instructions for Continued Airworthiness N8- EC135 V2-HV2 NUM E-A-07 | The system does not replace standard maintenance records and pilot reporting. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or rep (...) | 12.03.2013 |
| 10035287 | ISEI | | REV. 5 | | AS 350 B2, AS 350 B3 EC 130 B4 | Installation of Helicom V2 NUM Revision 1 introducing alternate positions for antennas Revision 2 correcting minor discrepancies in associated documentation Revision 3 Addition of "Instruction for Continued Airworthiness No N8-AS350 HV2-E-F-02 Rev 1", an (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Approval Change file No N1-AS350-HV2-E-F-02 Installation Manual No N7-AS350-HV2-E-F-02 Instruction for Continued Airworthiness No N8-AS350-HV2-E-F-02 Rev 1 | For AS350B2 model this STC is applicable only to aircraft equipped with VEMD system. The system does not replace standard maintenance records and pilot reporting. Prior to installation of this modification it must be determined that the interrelationship (...) | 03.09.2013 |
| 10038671 | ISEI | | | | EC 225 LP | Installation of the Helicom V2 NUM Embedded Equipment | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | N9-EC225 NUM-HV2 NUM Edit A Revision 02 dated 15 December 2011 - Flight Manual Supplement or later revisions of the above listed documents approved by EASA N8-EC225 NUM-HV2 NUM Edit A revision 02 dated 15 December 2011 - Instructions for Continued Airwor (...) | None | 07.03.2012 |
| 10039479 | ISEI | | | | AS 355 F, AS 365 F1, AS 365 F2 AS 355 N | Installation of: HELICOM V2 ANA ON PANEL Installation of a recorder and transmitter of flight-and-system parameters. The modification is defined in Document ISEI change N3-AS355 ANA-HV2 ANA ON PANEL-DESIGN CHANGE-E-A-04. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 27 Initial Issue for the change. Special Condition for Lithium Battery Installation Th (...) | N9- AS355 ANA-HV2 ANA ON PANEL -Edit A Revision 04 - Flight Manual Supplement - or later revisions of the above listed documents approved by EASA N2- AS355 ANA-HV2 ANA ON PANEL Edit A Revision 04 - Certification Program - N3- AS355 ANA-HV2 ANA ON PANEL Ed (...) | None | 08.05.2012 |
| 10044552 | ISEI | | | | MD500 N | Installation of Helicom V2 ANA Embedded UMS Equipment. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): ISEI MD369/500 (Generic) CRIF-01 "Lithium Battery Installations" dated 17 April 2013. The following FA (...) | Certification programme: N2-MD500-HV2-ANA (E-A-03) ICA: N8-MD500-HV2 ANA (E-A-03) or later revisions of the above listed documents approved by EASA | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 24.04.2013 |
| 10046224 | ISEI | | | | AS365 N2, AS365 N3 SA365 N, SA365 N1 | Installation of the Helicom V2 ANA embedded Equipment. The Helicom V2 ANA consists of a recording/transmitter equipment of flight and system's parameters. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI F-01 Lithium Battery, dated 20/02/2013. The requirements for environmental protection and the associated certified noise (...) | Design Change Description N3-AS365 ANA-HV2-ANA-E-A-06 dated 05/08/2013. Installation Manual N7-AS365 ANA-AS365-HV2-ANA-E-A-06 dated 05/08/2013. Maintenance Manual Supplement N8-AS365 ANA-AS365-ANA-E-A-06 dated 05/08/2013. | The change does not replace standard maintenance records and pilot reporting. Any broadcasting must be done in accordance with National regulations. Prior to installation of this design change it must be determined that the interrelationship between th (...) | 30.08.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|---------------------|----------------|-------------------------|--------------------------|--|---|---|--|---------------|
| 10016951 | ISOLAIR Inc. | EASA.IM.R.S.01346 | | FAA STC SH4796NM | BELL 206B | Installation of Isolair 3900-206 Spray System consisting of a fiberglass 130 gallon belly tank and spray booms for agricultural applications. | | 39206INS revision B dated 22 August 2006 or later approved revision # Installation Manual:3900-206 revision IR dated 14 August 2006 or later approved revision # Master Drawing List:3900ICA revision A dated 6 August 2006 or later approved revision # M (...) | Bell 206B High Skid Landing Gear must be installed. An aircraft with this STC installed will be operating in the Restricted Category. Essential crew required for the mission only. Vne is 90 KIAS. Spicing at airspeeds below 17 KIAS is prohibited; and Spra (...) | 06.03.2008 |
| 10039386 | ISOLAIR Inc. | | | FAA STC SR02813CH | ENSTROM 480 ENSTROM 480B | Title: Installation of 3900-480 Spray System. Description: The installation of the 3900-480 Spray System pertains to the installation of an agricultural spray system onto the Enstrom 480 and 480B rotorcraft. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement No. 3900-480FMS-EASA, Chapter 4 "Airworthiness Limitations" of the Instructions For Continued Airworthiness, ICA Manual No. 3900-480-EASA ICA, Rev. A dated 7th March 2012, or later revisions of the above listed document (...) | Restricted to agricultural use. | 02.05.2012 |
| 10041184 | ISOLAIR Inc. | EASA.IM.R.S.01018 | | | Data not found | Installation of Isolair spray system model 3900-350B2. | | - DECA Aviation Flight Manual Supplement No. 4849-02, Rev. 2, dated 22 August 2000, or later approved revisions. - DECA Aviation Modifications Summary List No. MS0006907, Issue 3, dated 5 April 2001, or later approved revisions. - Isolair Instructions 1 (...) | | 16.06.2009 |
| 10014423 | ISRAEL AEROSPACE INDUSTRIES, LTD. | EASA.IM.A.S.02039 | REV. 2 | CAA ISRAEL STC NO SA152 | 747-400 | Revision 1: Boeing 747-400SF conversion from PAX to Freighter CAAI STCSA152/EASA STC IM.A.S.02039. Revision 2: Increase of Max. Taxi Weight and Max. Take Off Weight from 873.000 / 870.000 to 878.000 / 875.000 Lbs | The Certification Basis for the original product remains applicable to this certificate/ approval. | Israel Civil Aviation Authority STC SA152 dated October 27, 2006. IAI MDL 366-00-00-A3673 Revision C dated October 22, 2006. IAI Doc. No. 366-00-00-A3663-481 issued May 26, 2011. Supplement to Airplane Flight Manual D6U10002.481 for IAI B747-400SFW (-&-&-)B (...) | none | 11.07.2011 |
| 10015354 | ISRAEL AEROSPACE INDUSTRIES, LTD. | EASA.IM.A.S.00632 | REV. 2 | | BOEING 767-200 | Conversion from Boeing 767-200 passenger aircraft to Boeing 767-200SF (Special Freighter) configuration, to enable for carriage of palletised cargo. Revision 1 to approved increased weight limitations | | - IAI MDL TR 368-00-00-94100 Revision AR or later EASA approved revision - AFM Supplement 368-00-00-91713 Revision 1 or later approved revision | 1. Supplementary Data Sheet TCDS.IM.A.S.00632. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 27.06.2007 |
| 10015542 | ISRAEL AEROSPACE INDUSTRIES, LTD. | EASA.IM.A.S.01426 | REV. 5 | | B737-300 | Conversion from Passenger configuration to Special Freighter (SF), Special Freighter with Rigid Barrier and to Quick Change (QC) configuration and approval of increased weight limitations. Summary of EASA STC Revisions: Original EASA approval: CAA UK STC (...) | EASA STC Certification Basis, as recorded in EASA CRI A-OI, dated 2007-11-02. The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged (...) | - Master Document List TR 365-2100-92100 Revision "BB", dated October 28, 2010. List of added converted airplanes: TR 365-00-00-99312 Revision "AE" dated September 18, 2007 or later revisions of the above listed documents approved by EASA | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. If the holder agrees (...) | 23.04.2012 |
| 10015911 | ISRAEL AEROSPACE INDUSTRIES, LTD. | EASA.IM.A.S.02040 | REV. 1 | | BOEING 747-400 | Conversion of Boeing 747-400 Aircraft from Combi to Special Freighter Configuration | | - Master Document List Number 366-00-00-A3550 Revision "New" or later approved revision, - Airplane Flight Manual Supplement Number 366-00-00A2911 Revision A or later approved revision. | 1. Supplementary Data Sheet TCDS.IM.A.A.02040 applies. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 27.06.2007 |
| 10016227 | ISRAEL AEROSPACE INDUSTRIES, LTD. | EASA.IM.A.S.02479 | REV. 5 | CAA ISRAEL STC SA160 | B737-400 | Major Change to STC 10016227 Revision 4 for the IAI 737-400 conversion to freighter aircraft. Update of the MDL TR 365-00-00-B4200 from issue N to Issue T. All design changes being of level 2 and approved by the CAAI, the working arrangement (EASA/CAAI 22Nov2004) applies. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | MDL TR 365-00-00-B4200 issue T or later revisions of the above listed documents approved by EASA | Not Applicable | 27.03.2012 |
| 10028430 | ISRAEL AEROSPACE INDUSTRIES, LTD. | P-EASA.IM.A.S.02477 | REV. 3 | CAAI STC SA218 | B767-300 | Conversion of Boeing B767-300 from Passenger Configuration to Special Freighter (BDSF) with Main Deck Class E Cargo Compartment. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate. For Significantly Changed Areas, Certification Basis is CS25 Amendment 2, including: Special Condi (...) | B767-300 Conversion from PAX to SF - Master Document List, 371-00-00-C0010, Revision E; Instructions for Continued Airworthiness, 371-00-00-C004, Revision New; AFM Supplement, 371-00-00-C0490, Revision 1. Additional data on the Certification Basis and (...) | An approved Cargo Loading System must be installed. | 20.11.2012 |
| 10015652 | ITOCHU AIRLEASE, INC. | EASA.IM.A.S.01627 | | ST10436SC | BOEING 767-300 | Reconfiguration of an existing three seating class interior with 199 passengers to a two seating class interior (economy & business seats) with 232 passengers. | | Future Jet Aerodesign LLC Master Data List 1610000 Revision F dated July 7, 2005 or later approved revision. AFM Supplement Document number 1610000-AFMS Rev IR dated July 15 2005 or later approved revision. | | 02.11.2005 |
| 10014840 | J & C ENGINEERING SERVICES Ltd. | EASA.A.S.03057 | | | BOEING B737-300 | TCAS II Installation | | - Report JC/ER/1011, Issue 5, dated 23th May 2007 - AFM Supplement, Report JC/FMS/005, Issue 1.1, dated 23-05-2007 - Instructions for continued airworthiness (ICA), TTF Aerospace Document TTF-05T026-18 revision new, dated February 13, 2006 or later FAA accepted revision. | Limited to installation on Boeing 737-300 MSN 28740. Compatibility of this design change with other previously approved modifications must be determined by the installer. | 29.05.2007 |
| 10014841 | J & C ENGINEERING SERVICES Ltd. | EASA.A.S.03058 | | | BOEING B737-300 | Installation of Driessen Galley No 1 | | - Report JC/ER/1012, Issue 3, dated 15th May 2007 | Limited to installation on Boeing 737-300 MSN 28740. Compatibility of this design change with other previously approved modifications must be determined by the installer. | 31.05.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|-------------------|----------------|----------------------------------|--|---|--|---|---|---------------|
| 10015174 | J & C ENGINEERING SERVICES Ltd. | EASA.A.S.03547 | REV 1 | | R2160, R2160D, R2160I, R2120U | Removal of VFR Limitation from AFM Approved Operation Section. | | Engineering Report JC/ER/1444 Issue 2.0 Flight Manual Supplements JC/FMS/1444 Issue 1.1, JC/FMS/1444-1 Issue 1.0, JC/FMS/1444-2 Issue 1.0, JC/FMS/1444-3 Issue 1.0 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 07.11.2008 |
| 10015175 | J & C ENGINEERING SERVICES Ltd. | EASA.A.S.03550 | | | BOEING 767-200 | Approval of Galley and Crew Rest Installation. | | J&C Engineering Services Modification JC/MOD/1428 Issue 1.1, 10/06/08. | | 12.06.2008 |
| 10044939 | J & C ENGINEERING SERVICES Ltd. | | | | PIPER AZTEC PA23-250 | Installation of a protruding camera on an approved fully provisioned location | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CAR 3, dated November 1, 1949, including Amendments 3-1 through 3-9, dated December 15, 1952, The requirements for avionics (...) | C/ER/M3125 Modification Sheet JC/FTR3125 Flight test report later revisions of the above listed documents approved by EASA, ... | As per flight manual supplement JC/FMS/3125. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 21.05.2013 |
| 10015356 | JAMCO AMERICA INC | EASA.IM.A.S.00647 | | FAA STC ST01372SE | BOEING 767-300 | New Seats and Interior Reconfiguration | | -FAA STC ST01372SE-Jamco America Inc Master Drawing List IA-103MDL Revision A dated March 4 2004 or later FAA approved revision | 1. Limitation on FAA STC in respect of concurrent installation of Thales IFE STC ST0162LA is not applicable. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". C (...) | 17.03.2005 |
| 10015496 | JAMCO AMERICA INC | EASA.IM.A.S.01258 | | FAA STC ST01121SE | BOEING 767-200 BOEING 767-300 BOEING 767-400ER | Installation of a new security flight deck door | | JAMCO America Inc. Master Drawing List IA-082DAL MDL Revision G dated March 18 2003 or later approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 21.06.2005 |
| 10016040 | JAMCO AMERICA INC | EASA.IM.A.S.02209 | REV. 1 | FAA STC ST01700SE | BOEING 747-400 | Removal of the strengthened flight deck door and modification and reinstallation of luggage bins on the upper deck in accordance with Jamco Aero Design & Engineering Master Drawing List JD-03AC02 Rev. F or later revisions of the above listed documents (...) | | 1. JD-03AC02 Rev. F Master Drawing List B747-400 Pax to Freighter upper deck reconfiguration. 2. B747-25-M05A Rev. F JADE Service Bulletin Pax to Freighter upper deck reconfiguration. | 1. This STC is limited to the following aircraft after incorporation of Boeing SB 747-00-2004 Passenger to Freighter conversion. RT501 Sn. 24061b, RT506 Sn.24066c, RT507 Sn. 24226d, RT509 Sn. 249752. This S (...) | 02.04.2007 |
| 10016307 | JAMCO AMERICA INC | EASA.IM.A.S.02604 | | FAA STC ST01459SE | BOEING 767-300 | Oxygen Module Installation. | | JAMCO Master Drawing List IA-122MDL, Revision C, 31/08/2004. | Reconfiguration of Passenger seats as per FAA STC SA7032NM-D must be installed concurrently. This FAA STC must be approved by EASA directly or by Grandfather Rights before installation on a EASA member state aircraft. | 10.01.2008 |
| 10016541 | JAMCO AMERICA INC | EASA.IM.A.S.02919 | | FAA STC ST00240SE ; P. GILLESPIE | BOEING 747-200 | Installation of Jamco Galley in accordance with Jamco America Inc. Master Drawing List IA-036, Rev. A dated September 1, 1995, or later approved revisions. | | None. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 16.01.2009 |
| 10032104 | JAMCO AMERICA INC | | | FAA STC ST01094SE | 747-400 | Installation of an interior reconfiguration of the main and upper decks in 747-400 airplanes in accordance with Jamco America, Inc. Master Drawing List no. IA-072MDL, Revision AD dated 19 September 2008 or later approved revision/Validation of FAA STC ST01094SE | The Certification Basis for the original product remains applicable to this certificate/ approval | or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 07.10.2010 |
| 10041373 | JAMCO AMERICA INC | | | FAA STC ST02107SE-D | B777-300 | Validation of FAA STC ST02107SE-D related to Interior Reconfiguration. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Requirements at a later amendment level than specified in TCDSEASA.IM.A.003 are listed (...) | Jamco America, Inc. Master Drawing List (MDL) mdl 1724-1, Revision B, dated November 3, 2011; Airplane Flight Manual Supplement, document AFM1724-1, Rev. 1, dated October 6, 2011; Supplemental Instruction for Continued Airworthiness (ICA), document number (...) | Approval of this change in type design is applicable to the above model aircraft only. | 11.09.2012 |
| 10041806 | JAMCO AMERICA INC | | | FAA STC ST01884SE-D | 747-400 | Installation of a galley modification on Boeing 747-400 series aircraft in accordance with Jamco America, Inc. Master Drawing List (MDL) 1419-1, Rev. E, dated May 25, 2012, or later FAA approved revision. Validation of FAA STC ST01884SE-D | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. § 25.853(a)(b)(c) Amdt 25-72 ; § 25.853(g) Amdt 25-116 ; § 25.853(h) Amdt 25-83 ; § 25.8 (...) | Jamco America, Inc. Master Drawing List 1419-1, Rev. E, dated May 25, 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | S/N 29262 | 17.10.2012 |
| 10042094 | JAMCO AMERICA INC | | | FAA STC ST01514SE | B747-400 | Lavatory and galley modification of 747-400 series airplanes in accordance with Jamco America, Inc. Master Data List IA-102MDL, Rev. D, dated March 15, 2005, or later FAA approved revision. Validation of FAA STC ST01514SE. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Jamco America, Inc. Master Data List IA-102MDL, Rev. D, dated March 15, 2005, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | FAA STC ST01492SE, interior reconfiguration of passenger seating, must be installed concurrently with this STC. | 07.11.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------|----------------|-----------------------|--|--|---|---|---|---------------|
| 10043568 | JAMCO AMERICA INC | | | FAA STC ST01494SE | 777-200, 777-200LR | Lavatory and Galley Modification | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | JAMCO America, Inc., Master Drawing List IA-118MDL, Revision C, dated December 29, 2004 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) # and/ or (...) | EASA validated version of FAA STC ST01493SE, interior reconfiguration of the passenger seating, must be installed concurrently with this STC. Prior to installation of this design change it must be determined that the interrelationship between this design and the original product is not affected. | 07.02.2013 |
| 10016454 | JAZZ AVIATION LP | EASA.IM.A.S.02800 | | TC STC SA03-70 | BOMBARDIER DHC-8-100 SERIES BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-300 SERIES | Cockpit Door Electronic Strike System. | | TTF Aerospace, LLC Master Drawing List (MDL), TTF-03T002-01, Rev. E, dated 13.06.2007 *) TTF Aerospace, LLC AFM Supplement TTF-03T002-15, Rev. New, dated 09.09.2003 *) TTF Aerospace, LLC Maintenance Manual Supplement TTF-03T002-12, Rev. E, date 07.05. (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. This (...) | 30.06.2008 |
| 10033847 | JCB AERO SA | | | | A320 | Installation of Hi-Low Table for VIP cabin interior | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List, ref. doc. A320ALL01-MDL-01, Revision B, dated 4th February 2011, Project Data List, ref. doc. A320ALL01-PDL-01, Revision A, dated 7th February 2011, Declaration of Compliance for Project A320ALL01, dated 11th February 2011 or (...) | none | 14.02.2011 |
| 10042732 | JCB AERO SA | | | | A330-223 | Galley G11 Installation* Installation of structural provisions for galley G11* Installation of Galley G11 (dry galley) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environment protection and the associated certificate noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/ approval. | Certification Plan A332-00229-02-CP-01, Rev. 1, dated 11st December 2012* Project Data List A332-00229-02-PDL-01, Rev. B, dated 14th December 2012* Declaration of Compliance A332-00229-02-DC-01, Initial Revision, dated 12th December 2012* Instructions (...) | The Pre-Mod status for this modification must be without seats or refurbishing. | 14.12.2012 |
| 10043374 | JCB AERO SA | | | | 737-800 | VIP Cabin Reconfiguration See Project Data List B738-35792-01-PDL-01 rev G. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. SCs or ESFs: CRI D-002 SC Isolated Compartments CRI D-012 SC Injury Criteria for Single (...) | Aircraft Flight Manual Supplement: B738-35792-01-AFMS-01 rev B Instructions for Continued Airworthiness: B738-35792-01-ICA-01 rev C Cabin Attendant Operating Manual: B738-35792-01-CAOM-01 rev C or later revisions of the above listed documents approved by EASA | S/N: 35792 Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 24.01.2013 |
| 10045092 | JCB AERO SA | | | | A330-223 | Staircase Removal Monuments Modification. Description - Summary : Some Cabin items removed (Galley G7, Lavatory L37, VCC, Staircase, Wardrobe), Lavatory L36 Modified & relocated, provisions for LDSCR (which is not installed). For further details also refer (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Project Data List A332-00324-01-PDL-01, Rev. 1, dated 11st December 2012* Project Data List A332-00324-01-CP-01, Issue DICA, A332-00324-01-ICA-01, Rev. 1 or later revisions of the above listed documents approved by EASA. | 1), Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 29.05.2013 |
| 10016010 | JCM AERODESIGN LIMITED | EASA.IM.A.S.02175 | | O-LSA06-268/D ISSUE 1 | BOMBARDIER DHC-8-311 | EMS Installation as per TCCA Limited Supplemental Type Certificate O-LSA06 268/D, Issue 1, dated 13 September 2006 | | Installation must be in accordance with Spectrum Aeromed Inc. Master Document List 6039-1A revision 1 or later TCCA approved revision Operation must be in accordance with Spectrum Aeromed Inc. Flight Manual Supplement 6039-1FMS at Revision 1 or later (...) | DHC-8-311, S/N: 216 | 20.12.2006 |
| 10016783 | JCM AERODESIGN LIMITED | EASA.IM.R.S.01124 | REV. 2 | TCCA STC SH02-2 | EUROCOPTER FRANCE AS 350 B3 | Installation of Optech Inc. Airborne Laser Terrain Mapper ALTM 1225 or ALTM 3033 in Eurocopter AS 350B1 and AS 350B2 rotorcraft (approved under EASA.IM.R.S.01124 # initial issue). Installation of Optech Inc. Airborne Laser Terrain Mapper ALTM 3100 in (...) | | Definition and installation: JCM Aerodesgin Limited dwg. No. JCM-013078 Revision E or later approved revisions. Inspection and maintenance: JCM Aerodesgin Limited Maintenance Manual Supplement No. 013078MMS Issue 5.1 or later approved revisions. Opera (...) | 1. This STC is not approved for AS350 B3 aircraft when equipped with Eurocopter Mod OP-3369 (2370 kg certified weight extension). 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and the original product is not affected. | 22.05.2008 |
| 10015264 | JET AVIATION AG | EASA.A.S.03726 | REV. 1 | | GULFSTREAM V | Iridium Antenna Installation a dual Iridium antenna has been installed on top of the fuselage located between Sta. 539.5 & Sta. 554.5. | | Iridium Antenna Installation In accordance with JET AVIATION Master Data List No.: BGLF553301-001 Rev 1 R, Dated 14th December 2008 or later approved revisions. Instructions for Continued Airworthiness ref BGLF553301-150-01 No AFMS has been generated for this modification. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 19.12.2008 |
| 10015626 | JET AVIATION AG | EASA.IM.A.S.01578 | REV. 2 | | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 | Activation of RAAS | | Master Data List No. BCL6134461-001, Rev. D of 12-01-2009 or later approved revision. AFM Supplement Bo. BCL6134462-201-01, Rev. A of 12-01-2009 or later revision. | None | 13.01.2009 |
| 10028110 | JET AVIATION AG | | | | GULFSTREAM GV | Mid Cabin Door | CS-25 The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Data List BGLF525001-001 Rev. A Airplane Flight Manual Supplement BGLF525001-201-01 Rev. R Instructions for Continued Airworthiness BGLF525001-150-01 Rev. A or later revisions of the above listed documents approved by EASA | None | 11.12.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|----------------|----------------|----------------------|-------------------------|---|---------------------|---|--|---------------|
| 10014526 | JET AVIATION AG, BASEL | EASA.A.S.02611 | | | B727-200 | ADI/ HIS & FMS upgrade by:removing both existing 329B-8x ADI 329Bremoving both existing 331A-9G HISremoving both LTN-SCCDUremoving both ALI 55B Radio Altimeter Indicatorsinstalling two new SATI E-ADI, p/n 195000-21installing two new ATI E-HIS, p/n 1 (...) | | Jet Aviation Declaration of Compliance for the project BACBA008.Master Data List, Doc. No. BACBA008-002, latest approved revision.AFMS for the 5-ATI EFI, Doc. No. BACBA008-201-01, Rev. B or latest approved revision.AFMS for the CMC-900 FMS, Doc. No. (...) | Reg.: A9C-BA, S/N 21824. Valid for this serial number only. | 04.09.2007 |
| 10014555 | JET AVIATION AG, BASEL | EASA.A.S.02653 | | | BAE 146/AVRO 146 SERIES | Cabin Interior Outfitting:Introducing a custom passenger cabin with individual interior and seating, VIP lavatory, In-flight Entertainment / Cabin ManagementSystem w/ tail and cockpit camera, Indium Satcom system, cabin power-outlet system and oxy (...) | | # Master Data List BRJE1267-001# Flight Manual Supplement BRJE1267-201-01# Instructions for Continued Airworthiness BRJE1267-150-01 | 1. Reg.: A6-RJK S/N: E1267.2. For limitations refer to Flight Manual Supplement BRJE1267-201-01. | 12.04.2007 |
| 10014576 | JET AVIATION AG, BASEL | EASA.A.S.02681 | | | BOEING 727-200 | Cabin interior modification:According to the description inthe Certification Plan, Doc. No. BACBA008-1. | | Declaration of Compliance dated 14 March 2007.Master Data List Doc. No. BACBA008-001.AFM Supplement Doc. No. BACBA008-201-12. | Reg.: A9C-BA, S/N: 218424Approval limited to S/N 21824 only. | 22.03.2007 |
| 10014577 | JET AVIATION AG, BASEL | EASA.A.S.02682 | | | B727-200 | Stretcher & Medical Equipment installation:According to thedescription in the Certification Plan, Doc. No. BACBA008-3 | | Declaration of Compliance dated 14 March 2007.Master Data List Doc. No. BACBA008-003.AFM Supplement Doc. No. BACBA008-201-12. | Reg.: A9C-BA, S/N: 21824Approval limited to S/N 21824 only. | 22.03.2007 |
| 10014613 | JET AVIATION AG, BASEL | EASA.A.S.02740 | | | BOEING 737-100 SERIES | Cabin ModificationGalley installation in the LH FWD Entrance AreaHigh Cabinet installation in the FWD RH Cabin | | - AFMS BPKAZ003-201-08- Instructions for Continued Airworthiness BPKAZ003-150-08 | 1. Approval valid only for s/n 32774 with FAA-STC No. ST09981SC installed.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other ai (...) | 16.11.2006 |
| 10014630 | JET AVIATION AG, BASEL | EASA.A.S.02770 | | | AIRBUS A319-115 | Cabin interior outfittingsee Master Data List BA3192949-001 | | Flight Manual Supplement BJC02949-201-01 | Limited to S/N 2949. | 19.07.2007 |
| 10014644 | JET AVIATION AG, BASEL | EASA.A.S.02796 | | | AIRBUS A319-100 | Installation Of Additional Water Tank | | Declaration of Compliance, Document Number BVCCJ003-001, Revision A, dated 16th April 2007, or later EASA approved revision.Master Data List, Document Number BVCCJ003-001, Revision A, dated 16th April 2007, or later EASA approved revision.Instructo (...) | | 24.04.2007 |
| 10014652 | JET AVIATION AG, BASEL | EASA.A.S.02807 | | | MYSTERE FALCON 900 | Wireless Access Service Point (WASP) | | - Master Data List, Doc. No. BCD01023-001 Revision A, dated 28.09.2006; - AFMS, Doc. No. BCD01023-201-13 Revision IR, dated 28.09.2006. | 1. Applicable for S/N 109 only.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 29.09.2006 |
| 10014659 | JET AVIATION AG, BASEL | EASA.A.S.02816 | | | BOEING 737-700 IGW | Wireless LAN (WLAN) Activation | | Master Data List BB73723261-001 Revision IR or later approved revisions | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 15.11.2006 |
| 10014691 | JET AVIATION AG, BASEL | EASA.A.S.02855 | REV. 1 | | A319 | Cabin interior outfitting | | Master Data List BCJ02921-001-REV.B, dated 25.08.2009Airplane Flight Manual Supplement BCJ02921-201-01-REV.IRInstructions for Continued Airworthiness BCJ02921-150-0-REV.IRor Intervisions of the above listed documents approved by EASA | Limited to Reg. P4-VNL, S/N 2921 | 02.11.2010 |
| 10014738 | JET AVIATION AG, BASEL | EASA.A.S.02921 | | | MYSTERE FALCON 50 | MK-V EGPWS upgrade. | | # Master Data List Doc No: BFA5034462-001.# AFMS Doc No: BFA5034462-201-01. | 1. Required Avionics Configuration:# Collins ProLine II# Collins EFIS 84/85/86# Collins WXP 8502. This STC is approved only for the product configuration as defined inthe approved design data referred to in the paragraph "Description". Co (...) | 04.01.2007 |
| 10014750 | JET AVIATION AG, BASEL | EASA.A.S.02935 | | | BOEING 727-200 | Activation of RAAS on MK-V or MK-VII EGPWS with Softwareversion -218-218 or higher up to -226-226 | | Declaration of Compliance dated 21.02.2007Master Data List No. BB72734461-001AFM Supplement No. BB72734461-201-01 | S/N 21824 as initial installationSTC Valid for B727-200 Series with EGPWS MK-V or MK-VII with Softwareversion -218-218or higher up to -226-226 | 14.03.2007 |
| 10014821 | JET AVIATION AG, BASEL | EASA.A.S.03032 | REV. 1 | | A330-202 | Cabin Interior Outfitting | | -Certification Plan, ref. doc. No. B3300605, Rev. B, dated 05.11.2008-Master Document List, ref. No. B3300605-001, Rev.IR, dated 03.11.2008-Airplane Flight Manual Supplement, ref. doc. No. B3300605-201-01, Rev.IR, dated 22.10.2008-Applicant's Declar (...) | None | 02.11.2010 |
| 10014859 | JET AVIATION AG, BASEL | EASA.A.S.03083 | | | BOMBARDIER BD-700-1A10 | W-LAN ActivationBy installing:A swivel dipole antenna in the cabin, a WLAN ON/OFF switch and new wiring between the router, antenna and the HD-710. | | # Declaration of Compliance, Doc. No. BGLX23261.# Master Data List, Doc. No. BGLX23261-001, Rev. A or latest approved revision.# AFMS, Doc. No. BGLX23261-201-01, Rev. IR or latest approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 06.11.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|----------------|----------------|----------------------|------------------------|--|---------------------|--|---|---------------|
| 10014868 | JET AVIATION AG, BASEL | EASA.A.S.03092 | | | AIRBUS A319-115 | Wireless Local Area Network (WLAN) Activation. The modification includes the following: The antenna will be connected and the WLAN will be activated. A "WLAN OFF" switch will be installed to switch the Wireless LAN functionality 'on' and 'off'. | | # Master Data List, ref. BA31923261-001, Rev. IR, dated 08.08.07# Certification Plan, ref. BA31923261, Rev. E, dated 31.07.07# Certification Compliance Sheet, ref. BA31923261-102-01, Rev. A, dated 01.08.07# Airplane Flight Manual Supplement (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 13.08.2007 |
| 10014888 | JET AVIATION AG, BASEL | EASA.A.S.03117 | | | AIRBUS A319-115 | Water Tanks Installation/The modification includes the following: Installation of two additional water tanks in the AFT lower lobe area between frame FR62 and frame FR65 on the LH and RH side. | | # Master Data List, ref. BA31938121-001, Rev. IR, dated 10.09.07# Certification Plan, ref. CP-BA31938121, Rev. C, dated 16.08.07# Certification Compliance Sheet, ref. BA31938121-102-01, Rev. C, dated 16.08.07# Instruction for Continued Ai (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 14.09.2007 |
| 10014904 | JET AVIATION AG, BASEL | EASA.A.S.03139 | | | GULFSTREAM III | Super FMS Installation by: Removing the existing dual UNS-1Msp Flight Management System (FMS) and install a new dual UNS-1L Super Flight Management System (FMS) a DTU-100 Data Transfer Unit. | | Master Data List, Doc No. BTAAW028-002, Rev A or latest approved revision AFMS, Doc NO BTAAW028-201-07, Rev IR or latest approved revision. | Reg.: T.J-AAW S/N 486 Valid for this S/N only. | 05.03.2008 |
| 10014915 | JET AVIATION AG, BASEL | EASA.A.S.03160 | | | GV | Installation of bulkheads and door in the mid cabin. | | Flight Manual Supplement doc no. GVCES002-201-02. | 1. S/N: 669.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness (...) | 19.10.2007 |
| 10014977 | JET AVIATION AG, BASEL | EASA.A.S.03265 | | | MYSTERE FALCON 50 | Triple FMS Upgrade to UNS-1F | | # Master Data List BOHIT039-002 Rev. E, dated 07th December 2007.# Airplane Flight Manual (AFM) Supplement BOHIT039-201-13, Rev. IR, dated 14th November 2007# Instructions for Continued Airworthiness BOHIT039-150-13, Revision B, dated 09th (...) | 1. Mystere Falcon MF 50 Serial Number 222See:# Airplane Flight Manual (AFM) Supplement BOHIT039-201-13, Rev. IR, dated 14th November 2007.# Instructions for Continued Airworthiness BOHIT039-150-13, Revision B, dated 09th October 2007, Or lat (...) | 17.12.2007 |
| 10014978 | JET AVIATION AG, BASEL | EASA.A.S.03267 | | | FALCON MF 50 MYSTERE | Interior Modifications. | | Master Data List BOHIT039-001 Rev. D Operation Instructions BOHIT039-202-11 Rev. IR Instructions for Continued Airworthiness BOHIT039-150-11 Revision A Engineering Order BOHIT039-250000 Rev. A OR any later approved revisions of these documents. | Mystere Falcon MF 50 Serial Number 222See: Operations Instructions BOHIT039-202-11 Rev. IR Instructions for Continued Airworthiness BOHIT039-150-11 Revision A | 17.10.2007 |
| 10015000 | JET AVIATION AG, BASEL | EASA.A.S.03300 | | | BOEING 767 | Activation of RAAS. | | Master Document List Doc. No.: BB7673446-001, AFM supplement Doc. No.: B76734461-210-01 | | 14.09.2007 |
| 10015019 | JET AVIATION AG, BASEL | EASA.A.S.03325 | | | AIRBUS A330 | Tailwind 550/560 Antenna and Rack Installation The modification includes the following:- Installation of a radome with mechanical antenna located top of the fuselage- Installation of a system rack inside the cabin, above the headliner | | ¿ Master Data List, ref. Doc. No. BA33053201-001, Rev. D, dated 27th June 2008¿ Certification Compliance Sheet, ref. Doc. No. BA33053201-102-01, Rev. A, dated 6th June 2008¿ Instruction for Continued Airworthiness, ref. Doc. No. BA33053201 (...) | Applicable A330-220/-300 MSN#s are listed in ref. doc. No. BA3305320-123, #Effectivity Substantiation Report#, Rev. IR, dated 23.04.08 | 03.07.2008 |
| 10015042 | JET AVIATION AG, BASEL | EASA.A.S.03358 | | | GULFSTREAM GV-SP G550 | Wireless LAN (WLAN) Activation | | Master Drawing List BGLF523261-001 Rev. AAFMS BGLF523261-201-01 Rev. IR | | 11.03.2008 |
| 10015077 | JET AVIATION AG, BASEL | EASA.A.S.03408 | | | GULFSTREAM III | Iridium SATCOM Installation consisting of-1 ICS-200 Iridium Transceiver- 1CS Iridium Antenna- 1 Sigma 7 handset- 2 Cordless Handset | | Master Data List, Doc. No. BTAAW028-001, Rev. A or latest approved revision, Instructions for Continued Airworthiness, Doc No. BTAAW028-150-09, Rev. Or latest approved revision. | Reg. T.J-AAW S/N 486 Valid for this S/N only. | 05.03.2008 |
| 10015090 | JET AVIATION AG, BASEL | EASA.A.S.03427 | | | BOMBARDIER CL-600-2B16 | Iridium Antenna Installation Aircell ST2100. | | Instruction for Continued Airworthiness, JBSL Doc. No. BCL6153301-150-01. | Reg.: ZS-SGC S/N: 5070 | 28.03.2008 |
| 10015167 | JET AVIATION AG, BASEL | EASA.A.S.03537 | | | MYSTERE FALCON 20 | Installation of a Honeywell MK VIII Enhanced Ground Proximity Warning System (EGPWS). | | # Master Data List BDMET063-001, Revision B dated 19th June 2008.# Airplane Flight Manual (AFM) Supplement BDMET063-201-25, Rev. B, dated 19th June 2008# Instructions for Continued Airworthiness BDMET063-150-25, Revision IR, dated 01st May 20 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. This (...) | 24.06.2008 |
| 10015201 | JET AVIATION AG, BASEL | EASA.A.S.03598 | | | GULFSTREAM IV | Wireless LAN (WLAN) Activation. | | Master Data List Doc. No. BGLF423261-001 AFM Supplement Doc. No. BGLF423261-201-01 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 28.08.2008 |
| 10015206 | JET AVIATION AG, BASEL | EASA.A.S.03619 | | | AIRBUS A319-133 | Wireless Local Area Network (WLAN) Activation | | ¿ Master Data List, ref. Doc. BA31923262-001, Rev. A, dated 13.08.08¿ Certification Compliance Sheet, ref. Doc. BA31923262-102-01, Rev. A, dated 12.08.08¿ Airplane Flight Manual Supplement, ref. BA31923262-201-01, Rev. IR, EASA approval date (...) | None | 13.08.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------|----------------|----------------------|---|---|---------------------|---|---|---------------|
| 10015228 | JET AVIATION AG, BASEL | EASA.A.S.03645 | | | BOMBARDIER CL-600-2B16 | Aircell Iridium Antenna P/N.015-10306-001 Installation | | Jet Aviation Declaration of Compliance dated 12 August 2008; Jet Aviation Master Data List No. B7REPCUS1044-001, Revision A dated 12 August 2008 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 13.08.2008 |
| 10015289 | JET AVIATION AG, BASEL | EASA.A.S.03824 | | | GULFSTREAM III | Activation of the Runway Awareness and Advisory System (RAAS) EGPWC upgrade to software level 228-228 and activation of the RAAS | | Installation in accordance with Jet Aviation Engineering Order BGLF334461 and Master Drawing List Doc No BGLF334461-001 Rev IR or later approved revisions. Operation in accordance with Jet Aviation Airplane Flight Manual Supplement ref BGLF334461-201-01 Rev IR or later approved revisions. | None | 11.05.2009 |
| 10015296 | JET AVIATION AG, BASEL | EASA.A.S.03844 | REV 2 | FOCA STC Z25-20-77 | BOEING 737-700 IGW | CTT Zonal dryer activation/Replacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via Swiss FOCA STC Z 25-20-77 due to the potential for combustion of contami (...) | | CTT SB 2001-006-21-01-AD Initial EASA AD 2006-0140-E or later approved revisions; Master Data List BBU32627-001 Rev. C dated 15.01.2009 or later approved revision | S/N 32627 | 21.04.2009 |
| 10015304 | JET AVIATION AG, BASEL | EASA.A.S.03886 | | | FALCON 900 | Activation of the Runway Awareness and Advisory System (RAAS) i.a.w. Jet Aviation Master Data List BF90034461-001 | | -Jet Aviation Master Data List BF90034461-001 Revision A dated 19th May 2009; -Jet Aviation Aeroplane Flight Manual Supplement BF90034461-201-01 Revision IR; -Jet Aviation Engineering Order (installation) BF90034461-344600 Revision IR or later EASA Approved revisions of these documents. | 1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2) Th (...) | 22.05.2009 |
| 10015444 | JET AVIATION AG, BASEL | EASA.IM.A.S.01185 | | - | AIRBUS A319-115 | Cabin interior outfitting | | Flight Manual Supplement BCJ02507-201-01 | Approval valid for S/N 2507 only. The aircraft is certified for private use only. It is prohibited to operate the aircraft for hire or offered for common carriage. The aircraft is a NO SMOKING aircraft. For other limitations, see flight manual supplement BCJ02507-201-01 | 20.12.2005 |
| 10015445 | JET AVIATION AG, BASEL | EASA.IM.A.S.01186 | | - | A320-232 | Cabin Modification (Final Completion of VIP Configuration) | | Flight Manual Supplement Doc. No BCJ02566-201-01 | 1) Modification limited to Aircraft S/N 25662. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 21.12.2005 |
| 10015446 | JET AVIATION AG, BASEL | EASA.IM.A.S.01187 | | - | LEARJET 31 LEARJET 31A LEARJET 35 LEARJET 35A LEARJET 36 LEARJET 36A LEARJET 55 LEARJET 55B LEARJET 55C LEARJET 60 | Activation of the Runway Awareness and Advisory System (RAAS). | | Master Data List Doc No BLJ3134461-001 FOCA approved AFM Supplement Document No BLJ3134461-201-01 Rev IR | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 30.05.2005 |
| 10015463 | JET AVIATION AG, BASEL | EASA.IM.A.S.01210 | | - | BOMBARDIER CL-600-1A11 (CL-600) | Flight Management System (FMS) Upgrade & Terrain Awareness Warning System (TAWS) Installation | | Jet Aviation Master Data List Document Number BOHET004-001 Rev B dated July 14 2005 or later approved revision; Jet Aviation AFM Supplement BOHET004-201-01 Rev B & BOHET004-201-04 Rev B or later approved revisions | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.07.2005 |
| 10015548 | JET AVIATION AG, BASEL | EASA.IM.A.S.01439 | | - | MYSTERE FALCON 50 | Installation of a Honeywell MK VII EGPWS | | Master Data List Doc. No.BFA5034461-001 AFM Supplement Doc.No.BFA5034461-201-01 | none | 15.11.2005 |
| 10015607 | JET AVIATION AG, BASEL | EASA.IM.A.S.01543 | | - | MYSTERE FALCON 50 | GPWS upgrade to EGPWS | | Master Data List Doc. No. BEHHS003-001 Revision A or later approved revisions; AFM Supplement Doc. No. BEHHS003-201-01 Revision IR or later approved revisions; ICA Doc. No. BEHHS003-150-01 Revision IR or later revisions. | This STC is limited to Falcon 50 S/N 204. | 17.08.2005 |
| 10015661 | JET AVIATION AG, BASEL | EASA.IM.A.S.01638 | REV 1 | | FALCON 900 | 1. FMZ-2000 Software Upgrade; 2. Application of Honeywell Service Bulletin No. 7018879-34-A6060. | | 1. Master Data List No. BF90034621 Rev. K or later approved versions; 2. AFM Supplement DOC No BF90034621-201-01, Rev. A. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 22.08.2008 |
| 10015689 | JET AVIATION AG, BASEL | EASA.IM.A.S.01674 | | - | GULFSTREAM IV | Installation of a L3 F2100 Solid State Flight Data Record (FDR) and Teledyne Mini Flight Data Acquisition Unit | | Master Data List Doc. GAHHH009-001 Rev. B | 1. Limited to Gulfstream IV (S/N. 1011 - A6-HH) | 15.11.2005 |
| 10015722 | JET AVIATION AG, BASEL | EASA.IM.A.S.01724 | | - | AIRBUS A320-232 | Cabin Interior Outfitting See Master Data List B3202403-001 | | Flight Manual Supplement B3202403201-01 Instructions for Continued Airworthiness B3202403-150-01 | 1. Approval valid only for MSN 2403.2. The aircraft is certified for private use only, not to be operated for commercial air transport. 3. The aircraft is a No Smoking aircraft. 4. The W-LAN System is limited to Cabin W-LAN, which is not int. (...) | 06.09.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------|----------------|----------------------|---|--|--|--|--|--------------------------|
| 10015729 | JET AVIATION AG, BASEL | EASA.IM.A.S.01748 | | | BOEING 737-700 IGW | Cabin Modification - Aft Lounge Modification - Replacement of IFE/CMS and Installation of Airshow 4000 | | Master Data List BVBRM024-001 Operation Manual Supplement BVBRM024-202-01 Instruction for Continued Airworthiness BVBRM024-150-01 | Approval valid only for MSN 28976 with FAA STC No. ST09106AC-D installed | 15.11.2006 |
| 10015752 | JET AVIATION AG, BASEL | EASA.IM.A.S.01785 | | - | BOEING 737-800 | Installation of modified customized seats | | Flight Manual Supplement Doc No BBJ33473-201-06 | 1. Modification limited to Aircraft S/N 334732. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 30.01.2006 |
| 10015769 | JET AVIATION AG, BASEL | EASA.IM.A.S.01808 | | - | FALCON 2000 | Cabin Reconfiguration | | Flight Manual Supplement Doc. No BOCKN019-201-01 | Modification limited to Aircraft S/N 076 | 09.02.2006 |
| 10015774 | JET AVIATION AG, BASEL | EASA.IM.A.S.01818 | | - | AIRBUS A320-111 AIRBUS A320-211 AIRBUS A320-212 AIRBUS A320-214 AIRBUS A320-231 AIRBUS A320-232 AIRBUS A320-233 | CAIR TM System Installation | | Instructions for Continued Airworthiness BA32021701-150-01 | The aircraft maximum number of passengers is reduced by 7 (seven) passengers. | 30.08.2006 |
| 10015775 | JET AVIATION AG, BASEL | EASA.IM.A.S.01819 | REV 1 | | AIRBUS A319-111 AIRBUS A319-112 AIRBUS A319-113 AIRBUS A319-114 AIRBUS A319-115 AIRBUS A319-131 AIRBUS A319-132 AIRBUS A319-133 AIRBUS A320-111 AIRBUS A320-211 AIRBUS A320-212 AIRBUS A320-214 AIRBUS A320-231 AIRBUS A320-232 AIRBUS A320-233 | Camera Installation See Master Data List BA32023391-001 | | Instructions for Continued Airworthiness BA32023391-150-01 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 10.09.2007 |
| 10015779 | JET AVIATION AG, BASEL | EASA.IM.A.S.01825 | | - | GULFSTREAM 1159 (G-II) | Weather Radar Installation | | Declaration of Compliance Project BTAAI008 Master Data List BTAAI008-001 | -- | 05.05.2006 |
| 10029053 | JET AVIATION AG, BASEL | EASA.A.S.03258 | | | BOEING 737-700 IGW MYSTERE FALCON 50 (MF50) | Runway Awareness and Advisory System (RAAS). Smart Runway Activation i.a.w. Jet Aviation Master Data List BF5EX34461-001 Revision B or later EASA Approved revisions. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. In addition compliance has been shown i.a.w. EASA CRI F-01 Issue 02 "Activation of the (...) | See: Jet Aviation Master Data List BF5EX34461-001 Revision B and Jet Aviation Aeroplane Flight Manual Supplement BF5EX34461-201-01 Revision IR Jet Aviation Engineering Order (Installation) BF5EX34461-344600 Revision B or later revisions of the above listed documents approved by EASA | ... This Jet Aviation Supplemental Type Certificate is applicable on Mystere Falcon 50 EX (F50EX) version airplanes with an existing and design approved Honeywell EGPWS Mark V installation. Caution alerts, terrain display text messages, take-off flap (...) | 15.07.2009 30.03.2010 |
| 10029553 | JET AVIATION AG, BASEL | P-EASA.A.S.03788 | REV. 1 | | B737-800 | EMS SATCOM AMT-3800 Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CRI H-01 Issue 2 /Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane (...) | Jet Aviation Master Data List No.: BB73723263-001, Rev. C, 07.06.2010 or later revisions of the above listed documents approved by EASA | Not Applicable | 12.07.2010 |
| 10029554 | JET AVIATION AG, BASEL | P-EASA.A.S.03517 | REV. 1 | | AIRBUS A319-115 | Airbus A319-115 "BCJ03513" VIP Cabin Interior Completion Revision 1: Update of the Instructions for Continued Airworthiness | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval Certification Review Item (CRI) G-4301-001 Issue 2 from 04 Nov. 2009 EASA Supplemental (...) | Airplane Flight Manual Supplement BCJ03513-201-01 Revision IR from 15 March 2010 Instructions for Continued Airworthiness BCJ03513-150-01 Revision C from 02 Aug. 2010 Master Data List BCJ03513-001 Revision C from 03 Aug. 2010 Declaration of Compliance CAF 3607 from 03 Aug. 2010 or later revisions of the | Not Applicable | 04.08.2010 |
| 10029953 | JET AVIATION AG, BASEL | P-EASA.A.S.03531 | REV. 1 | | B737-800 | VIP Cabin Interior Outfitting | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CRI A-01 Issue 4 for this project dated 09.07.2010. The requirements for environmental p (...) | Jet Aviation Documents - Master Data List: BBJ37545-001 Rev.D dated 09. July 2010 - AFM Supplement: BBJ37545-201 Rev. E - Inst. for Continued Airworthiness: BBJ37545-150-01 Rev. I for later revisions of the above listed documents approved by EASA | Limited to: S/N 37545 | 12.07.2010 |
| 10030964 | JET AVIATION AG, BASEL | P-EASA.A.S.03518 | REV. 1 | | A319-115 | Airbus A319-115 "BCJ03632" VIP Cabin Interior Completion Revision 1: Update of the Instructions for Continued Airworthiness | EASA TCDS A.064 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval Certification Review Item (CRI) G-4301-001 Issue 2 from 04 Nov. 2009 EASA (...) | Airplane Flight Manual Supplement BCJ03632-201-01 Revision A dated 09 July 2010. Instructions for Continued Airworthiness BCJ03632-150-01 Revision C dated 02 Aug. 2010. Master Data List BCJ03632-001 Revision D dated 03 Aug. 2010. Declaration of Compliance (...) | None | 04.08.2010 |
| 10032331 | JET AVIATION AG, BASEL | | | | B747-400 | Replacement of a dual Allied Signal High Frequency (HF) Radio by dual Rockwell Collins Air Transport (RCAT) HFS-900D HF Radio equipment (ARINC753) with HF Data Link activation in accordance with Jet Aviation Master Data List Doc. No. B7428343-006 Rev (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. | none | none | 27.10.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|------------------|----------------|----------------------|--|---|---|---|---|---------------|
| 10034437 | JET AVIATION AG, BASEL | | | | A319-100 | GSM Securaplane Antenna Installation | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI-H-01 'Enhanced Airworthiness Programme for Aeropla (...) | - Master Data List ref. doc. BA31953001-001, Rev. C (>-<-) dated 30th March 2011 - Certification Plan ref. doc. BA31953001, Rev. A, dated 31st March 2011 - Certification Compliance Sheet, ref. doc. BA31953001-102-01, Rev. A, dated 29th March 2011 - Decla (...) | None | 05.04.2011 |
| 10034649 | JET AVIATION AG, BASEL | | | | MF900 (EXCEPTED F900C) | NZ2000 6.1 upgrade with CD820 CDU replacement. Existing dual triple Honeywell NZ2000 Flight Management System is upgraded as follows: - NCU NZ2000 FMS 1&2 version NZ5.2 or later is replaced with version NZ6.1. CD800/CD810 1&2 CDU replaced with CD820. No change in sensors and capabilities. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List Doc. No. BF90034623-001 Rev. B dated 01.02.2011 Instructions for Continued Airworthiness Doc. No. BF90034623-150-01 dated 20.12.2010 (Airplane Flight Manual Supplement Doc. No. BF90034623-201-01 Rev. IR dated 17.03.2011 or later revisions of the above listed documents approved by EASA. | As per AFMS Doc. No. BF90034623-201-01 Rev. IR dated 17.03.2011. | 18.04.2011 |
| 10035212 | JET AVIATION AG, BASEL | EASA.A.S.03807 | | | FALCON 900EX | Activation of the Runway Awareness and Advisory System (RAAS) i.a.w. Jet Aviation Master Data List BF9EX34461-001 | | #, Jet Aviation Master Data List BF9EX34461-001 Revision B, dated 06th July 2009#, Jet Aviation Aeroplane Flight Manual Supplement BF9EX34461-201-01 Revision IR#, Jet Aviation Engineering Order (Installation) BF9EX34461-34460/Revision IR Or later EASA Approved revisions of these documents. | 1., Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2., T (...) | 08.07.2009 |
| 10035406 | JET AVIATION AG, BASEL | | | | FALCON 2000 | Iridium antenna installation. Structural justification of the installation of an Aircell low profile dual patch antenna on top of the fuselage between frames 17 and 18. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List Doc. No. BOSIR001-002 Rev. A dated 16.06.2011 Certification Compliance Sheet Doc. No. BOSIR001-102-02 Rev. A dated 28.01.2011 Instructions for Continued Airworthiness Doc. No. BOSIR001-150-02 Rev. IR dated 09.06.2011 for later revisions of the above listed documents approved by EASA. | Restricted to S/N 173 This approval only covers the structural installation of the antenna. Installation and activation of any related Satcom system must be justified and covered by another suitable approval. | 21.06.2011 |
| 10037000 | JET AVIATION AG, BASEL | P-EASA.A.S.03714 | | | B747-400 | Installation of Cabin Interior Doors in accordance with Jet Aviation Master Data List BHHM1001-001 Rev. B, dated 19th October, 2011 for B747-400 S/N 28343 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI-D-01 Doors between passenger compartments CRI-D-02 Isolated Compartment | BHHM1001-201-02 Rev. IR - Airplane Flight Manual Supplement BHHM1001-150-02 Rev. IR - Instructions for Continued Airworthiness or later revisions of the above listed documents approved by EASA | This aircraft is not eligible for registration in a European country, because it does not comply with the JAA Special Conditions 1 and 2. | 24.10.2011 |
| 10037425 | JET AVIATION AG, BASEL | | | | FALCON 2000EX | Change consists in activating WLAN WiFi function of existing Satcom/LAN Router equipment. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI-F-2104-001 Means of Compliance "Wireless Local Area Network (WLAN) Activation" close (...) | Master Data List Doc. No. BF2TX23261-001 Rev. B dated 10.11.2011 Instructions for Continued Airworthiness Doc. No. BF2TX23261-150-01 Rev. IR dated 28.06.2011 Airplane Flight Manual Supplement Doc. No. BF2TX23261-201-01 Rev. IR dated 27.06.2011 or later r (...) | Aircraft must be fitted with Dassault Aviation M3196 that installs complete Aircel Axes II system, including WiFi antennas. | 25.11.2011 |
| 10037504 | JET AVIATION AG, BASEL | EASA.A.S.03322 | REV. 2 | | B737-700 | EMS SATCOM AMT-3800 Installation: Installation option between STA.727E to STA.787 and STA.727B to STA.727E (Major change to EASA STC No. EASA.A.S.03322 Rev.1) | JAR 25, Change 13 The Certification Basis for the original product remains applicable to this certificate/ approval. Derogation to latest amendment (Part 21A, 101) Do not materially contribute to safety. The certificated noise and/or emissions levels of (...) | ICA Instruction for Continued Airworthiness Doc. BB73723262-150-02/ALS mandatory structural inspection requirements item no. 4.1.1 & 4.1.2) or later revisions of the above listed documents approved by EASA. | None. | 02.12.2011 |
| 10038393 | JET AVIATION AG, BASEL | | | | A319-133 | VIP Cabin Interior Completion See Master Data List BCJ04151-001 | JAR 25, Amdt: 11 except as noted in TCDS EASA.A.064. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI-A-4301-006 EASA STC Certificate (...) | Airplane Flight Manual Supplement BCJ04151-201-01 Rev. B Instructions for Continued Airworthiness BCJ04151-150-01 IREWI Instructions for Continued Airworthiness BCJ04151-150-02 Rev. A or later revisions of the above listed documents approved by EASA. | Approval valid only for MSN 4151. The aircraft is certified for private use only, not to be operated for commercial air transport. The aircraft is a No Smoking aircraft. For other Limitations refer to Airplane Flight Manual Supplement BCJ04151-201-01 Rev. B or later revision approved by EASA. | 15.02.2012 |
| 10038721 | JET AVIATION AG, BASEL | | REV. 1 | | EMB-135BJ | Installation of 4 Place Ambulance Stretcher System/ VIP interior. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: SC-D-4116-001 issue 1, dated 25th November 2011; SC-D-4126-001 issue 2, dated 8th March (...) | Master Data List reference 2011-8-16, Revision A, dated 26th February 2012; Engineer Order reference 2011-8, Revision A, dated 26th February 2012; AFM Supplement, reference 2011-8-8, Revision A, dated 24th February 2012; or later revisions of the above listed documents approved by EASA. | None. | 18.04.2013 |
| 10039269 | JET AVIATION AG, BASEL | P-EASA.A.S.03779 | REV. 2 | | A340-600 | Airbus A340-600 'Eminence' VIP Cabin Interior Completion Revision 2 to update the Instruction of Continued Airworthiness and AFM | EASA TCDS A.015 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: SC-D-1302-002 Isolated Compartments SC-D-1304-002 Flight Crew Rest Compa (...) | Airplane Flight Manual Supplement B3400924-201-01 Revision H-Instructions for Continued Airworthiness B3400924-150-01 Revision C from 17 Oct. 2012; Master Data List B3400924-001 Revision H from 01 Nov. 2012; Jet Aviation Declaration of Compliance for p (...) | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product. | 06.02.2013 |
| 10039356 | JET AVIATION AG, BASEL | | | | A340-211, A340-212, A340-213, A340-311, A340-312, A340-313 | Camera Unit and Structural provision installation This STC installs one camera on the tail of the aircraft, watching forward. This covers the mechanical installation of the camera unit and the penetration of the pressure. | The Certification basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List, BA34031701-001 Issue A, dated 26.04.2012 Instructions for Continued Airworthiness, BA34031701-150-01 Issue A, dated 26.04.2012 or later revisions of the above listed documents approved by EASA. | None. | 30.04.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|------------------|----------------|----------------------|--|---|---|---|---|---------------|
| 10039357 | JET AVIATION AG, BASEL | | | | A340-211, A340-212, A340-213 A340-311, A340-312, A340-313 | Tailwind 550/560 antenna installation. This STC installs a tailwind SAT TV antenna and radome on top of the fuselage. | The certification basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Master Data List, BA34053203-001 Issue A, dated 26.04.2012. Instructions for Continued Airworthiness, BA34053203-150-01 Issue A, dated 26.04.2012. or later revisions of the above listed documents approved by EASA. | None. | 30.04.2012 |
| 10039436 | JET AVIATION AG, BASEL | P-EASA.A.S.03778 | REV. 2 | | A340-300 | Airbus A340-300 "Olbia" VIP Cabin Interior Completion. This STC installs a VIP cabin interior. The Revision 1 removed a limitation. The Revision 2 has introduced two configurations regarding the Crew Rest Compartment utilisation. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): D-2302-001, Isolated Compartments D-2304-001, Crew Rest Compartments D-2310-001, Glass Screens of Disp (...) | Master Data List, ref : B3400955-001 Revision H, dated 17 July 2013. Airplane Flight Manual Supplement, ref : B3400955-201-01 Revision E, dated 28 May 2013. Instructions for Continued Airworthiness B3400955-150-01 Revision A, dated 26 April 2012. Weight and (...) | Airstairs installation is not part of the approved configuration | 19.07.2013 |
| 10039715 | JET AVIATION AG, BASEL | | | | FALCON 2000 | WLAN activation. For WLAN working in the 2.4 GHz frequency band | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List, Doc. No. BF2TH23262-001, Rev. IR, 20.04.2012- AFMS, Doc. No. BF2TH23262-201-01, Rev. IR, 29.02.2012- ICA, Doc. No. BF2TH23262-150-01, Rev. IR, 27.02.2012. or later revisions of the above listed documents approved by EASA. | | 24.05.2012 |
| 10040166 | JET AVIATION AG, BASEL | | | | 737-800 | Iridium Antenna Structural Provisions between fuselage station BS 727C & 727D | JAR 25 Change 13. The Certification Basis for the original product remains applicable to this certificate/ approval. SCS or ESFs: Derogation to latest amendment (Part 21A.101). Do not materially contribute to safety. The certified noise and/ or emission (...) | Instructions for Continued Airworthiness Doc. No. BB73723266-150-01 or later revisions of the above listed documents approved by EASA | none | 26.06.2012 |
| 10040716 | JET AVIATION AG, BASEL | | | | A319-115 | VIP Cabin Interior Completion. See Master Data List BCJ03856-001. | JAR 25, Ch. 11 except as noted in TCDS EASA.A.064. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI A-4301-016 EASA STC Certification (...) | Airplane Flight Manual Supplement BCJ03856-201-01 Rev. C. Instructions for Continued Airworthiness BCJ03856-150-01 IR. EWIS Instructions for Continued Airworthiness BCJ03856-150-02 Rev. IR, or later revisions of the above listed documents approved by EASA. | Approval valid only for MSN 3856. The aircraft is certified for private use only, not to be operated for commercial air transport. The aircraft is a No Smoking aircraft. For other Limitations refer to Airplane Flight Manual Supplement BCJ03856-201-01 Rev. C or later revision approved by EASA. | 23.07.2012 |
| 10040780 | JET AVIATION AG, BASEL | | | | B737-800 | Forward VIP Lavatory reconfiguration- new shower system- relocation of handspray system- new vanity light system- new shower light system- new pax sign inside shower- adaptation of existing systems | JAR 25, Amdt: Change 13, except as noted in EASA TCDS IM.A.120. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI D-01 Shower Installa (...) | Instructions for continued airworthiness ref. B6AUH003-150-19, Master Data List B6AUH003-019, Flight attendant handbook ref. WH-04, or later revisions of the above listed documents approved by EASA. | S/N: 33473 (A6-AUH) | 26.07.2012 |
| 10040974 | JET AVIATION AG, BASEL | | REV. 1 | | A330-201, A330-202, A330-203 A330-223, A330-243 | EMS Satcom AMT-3800 antenna installation. This modification installs EMS Satcom AMT-3800 high gain antenna, Diplexer, Low Noise Amplifier, and associated structural provision. The revision has withdrawn the STC limitations following additional substantiations delivery. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List, ref : BA33023261-001, Revision E. Instructions for Continued Airworthiness, ref : BA33023261-150-01, Revision C. Weight & Balance Statement, ref : BA33023261-121-01, or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 23.01.2013 |
| 10041576 | JET AVIATION AG, BASEL | | REV. 1 | | FALCON 10 | Exchange to Universal Avionics FMS UNS-1LwOut: Universal Avionics FMS UNS-1Msp system: Universal Avionics FMS UNS-1Lw system | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List Doc. No. BSDYB004-004 Rev. C of 21.02.2013. Instructions for Continued Airworthiness Doc. No. BSDYB004-150-04 Rev. A of 24.01.2013. Airplane Flight Manual Supplement Doc. No. BSDYB004-201-04 Rev. A of 01.02.2013. Declaration of Compliance P (...) | S/N: 216, 224. The approval holder shall fulfil the obligations of Part 21, Point 21.A.109 or 21.A.451(b). Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously ins (...) | 25.02.2013 |
| 10041595 | JET AVIATION AG, BASEL | | REV. 1 | | B737-900ER | EMS SATCOM AMT-3800 Installation. | FAR 25 Amdt: 25-108. The Certification Basis for the original product remains applicable to this certificate/ approval. Do not materially contribute to safety. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Instructions for Continued Airworthiness BB73723265-06, IR or later revisions of the above listed documents approved by EASA. | 1) This STC has been assessed for compatibility for installation on aircrafts, which are equipped for lower cabin altitude LOC under FAA STC ST016975E. However, EASA has not yet approved the lower cabin altitude STC. Therefore, if such an aircraft is t (...) | 18.01.2013 |
| 10041604 | JET AVIATION AG, BASEL | | REV. 1 | | B737-900ER | VIP Cabin interior installation. Installation of table leaf adjacent to RH over-wing exit. See Master Data List BBJ37560-001 Rev. C (or later approved revision). | FAR 25, Amdt: 25-108. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI A-01 EASA STC Certification Basis SCS or ESFs: CRI C-01 Interf (...) | Airplane Flight Manual Supplement BBJ37560-201-01 Rev. D. Instructions for Continued Airworthiness BBJ37560-150-01 Rev. IR. EWIS Instructions for Continued Airworthiness BBJ37560-150-02 or later revisions of the above listed documents approved by EASA. | S/N: 37560 | 05.11.2012 |
| 10042418 | JET AVIATION AG, BASEL | | | | A319-115 | Cabin Interior Outfitting. See Master Data List BCJ04353-001 Rev. C | JAR 25 change 11 except for those paragraph listed in the TCDS EASA.A.064. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: SCS or ESFs: CR (...) | Aircraft Flight Manual Supplement: BCJ04353-201-01 Rev. E. Instruction for Continued Airworthiness: BCJ04353-150-01 Rev. A. EWIS Instructions for Continued Airworthiness: BCJ04353-150-02 Rev. A. Cabin Attendant Operating Manual AIY-04 Rev. C, or later re (...) | S/N: 4353 | 30.11.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|----------------------|--|---|--|---|--|---------------|
| 10042850 | JET AVIATION AG, BASEL | | REV. 1 | | GV-SP | MDL BDBMW002-001 Rev. IR | JAR-25 Change 15 and JAR AWO Change 2 with the exceptions as per EASATCDS IM.A.070The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product (...) | Rosen Aviation Remote Display STM 105478 Rev. Aor later revisions of the above listed documents approved by EASA | S/N: 5336 | 21.12.2012 |
| 10042977 | JET AVIATION AG, BASEL | EASA.IM.A.S.01246 | | | B737-700 | The installed CTT Zonal Comfort System - CAIR System consists of Zonal Dryer - (behind aft cargo compartment)Humidifier1 (FWD) - (LH Air Conditioning Mixing Bay)Humidifier 2 (AFT) - (RH Air Conditioning Mixing Bay)Humidifier Control Box -(Center Rac (...) | | As per Master Data list BADAS004-001 | 1. This STC is only valid for Serial Number 298582. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 14.07.2005 |
| 10043088 | JET AVIATION AG, BASEL | EASA.A.S.03597 | | | GULFSTREAM IV | FMS Upgrade Installation new EHSI 4000. NZ2000 SW 6.0 Installation, New CDU CD-820. | | - Master Data List, doc. no. GHIMY221-001, Rev. IR- AFMS (EHSI), doc. no. GHIMY221-201-06, Rev. IR- AFMS (FMS), doc. no. GHIMY221-201-14, Rev. IR- ICA (EHSI), doc. no. GHIMY221-150-06, Rev. IR- ICA (FMS), doc. no. GHIMY221-150-14, Rev. IR | Reg.: HB-IMY, S/N: 1084 | 18.05.2009 |
| 10043134 | JET AVIATION AG, BASEL | EASA.A.S.03457 | | | B747-400 | Interior Modifications | | - MDL Doc. No. BAHRM004-001, Rev. BAFMS Doc. No. BAHRM004-201-01, Rev. IROMM Doc. No. VB-06, Rev. B | 1. STC limited to Aircraft S/N 26903.2. For other Limitation refer to the AFMS BAHRM004-201-0.3. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modifications (...) | 02.06.2009 |
| 10043269 | JET AVIATION AG, BASEL | | | | MF50 (EX-VERSION) | WLAN activationElectro-Magnetic Compatibility demonstration of airplane's required systems immunity to worst case WLAN IEEE 802.11 a/b/g/n passengers carry on TPED transmissions. | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Data List, Doc. No. BFSLA001-002, Rev. IR, dated 11.01.2013- AFMS, Doc. No. BFSLA001-201-02, Rev. IR, dated 11.01.2013- ICA, Doc. No. BFSLA001-150-02, Rev. IR, dated 10.12.2012or later revisions of the above listed documents approved by EASA. | S/N: 348 F-GLSA | 17.01.2013 |
| 10046144 | JET AVIATION AG, BASEL | | | | B737-700 | Cabin Insulation Improvement as defined in BB73725801-500_REV_A_ProjectDescription and MDL BB73725801-001 Master Data List | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Engineering Order, Doc. No. BB73725801-258000, Rev. IR, 26.06.2013- Master Data List, Doc. No. MDL BB73725801-001, Rev. A, 08.08.2013or later revisions of the above listed documents approved by EASA. | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 21.08.2013 |
| 10014941 | JET AVIATION AG, ZURICH AIRPORT BRA | EASA.A.S.03200 | | | CESSNA 560 | EHS integration into Aircraft.By upgrading both existing RCZ-833E to RCZ-833K and integrated Mode S Transponder Module XS-852A to XS-852F, with the Honeywell Service Bulletin No. 7510700-23-6046. | | Master Data List, Doc. MDL-055, Rev. IR or latest approved Revision.Flight Manual Supplement, Doc. No. FMS-028, Rev. IR or latest approved revision.Instruction for Continued Airworthiness, Doc. No. ICA-021, Rev. IR or latest approved revision. | Ref.: \ S/N: Prerequisite equipment- IC-600 CDG, P/N 7017000-80162- IC-600 CDG, P/N 7017000-81162- RCZ-833E ICU, P/N 7510700-866- RM-850 RMU, P/N 7012100-825- GNS-XL FMS, P/N 18355-0101-0101For aircrafts equipped with a CAS-67A TCAS II-TPU 67-A, P (...) | 08.10.2007 |
| 10026896 | JET AVIATION AG, ZURICH AIRPORT BRA | P-EASA.A.S.03967 | | | HS 125 SERIES 700B | Cockpit Voice Recorder InstallationBy installing a new L3 FA2100 Cockpit Voice Recorder (CVR) | FAA CAR-4b, FAR 25 for the Change | - Master Document List, Doc. No. MDL-068, Rev. B or latest approved Revision - AFMS, Doc. No. FMS-041, Rev. IR or latest approved Revision - ICA, Doc. No. ICA-034, Rev. IR or latest approved Revision | None | 20.08.2009 |
| 10027723 | JET AVIATION AG, ZURICH AIRPORT BRA | | | | 525 / 525A / 525B | MINIQAR installation; By installing an Avionics miniQAR MK II and the MKIII according to the Installation Instructions OPN 009. | FAR 23.The Certification Basis for the original product remains applicable to this certificate/ approval.Derogation to latest amendment (Part 21A.101); non significant change.The certificated noise and/ or emissions levels of the original product are unchanged and (...) | - Master Document List, Doc. No. MDL-069-ICA, Doc. No. ICA-036or later revisions of the above listed documents approved by EASA. | S/N 525-0600 & ON, 525A-0001 & ON, 525B-0135 & ON. | 26.10.2009 |
| 10036985 | JET AVIATION AG, ZURICH AIRPORT BRA | | | | CESSNA 525 | Removed Equipment:-Aerosonic VSI 1 & 2-Skywatch 497 TCASI SystemNew installed equipment:-Honeywell TA/ VSI Indicator 1 & PS-576A XPR/TCAS Control HeadCAS-67A TCAS II System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.SCs or ESFs: Voluntary compliance to CS-23 First Issue for the changeDerogation to lat (...) | - Master Document List, Doc. No. MDL-075, Rev. B- ICA, Doc. No. ICA-040, Rev. A- AFMS EHS, Doc. No. FMS-042, Rev. IR or later revisions of the above listed documents approved by EASA | S/N: 525-0322 F-HCPB | 24.10.2011 |
| 10040249 | JET AVIATION AG, ZURICH AIRPORT BRA | | REV. 1 | | B200, B200C, B200GT, B200CGT B300, B300C | ADS-B-OUT ComplianceOut: Dual Collins TDR-94D Transponder Mode S P/N 822-9210-008In: Dual Collins TDR-94D Transponder Mode S P/N 622-9210-409 & 1 dual annunciator/switch ADS-B 1/ADS-B 2 inhibit | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.SCs or ESFs: CS 23 for the changeThe certificated noise and/ or emissions levels of th (...) | Master Document List MDL-100 Rev. B 12.06.2012Instructions for continued Airworthiness ICA-100 Rev. A of 02.05.2012AFM Supplement FMS-100 Rev. IR of 22.05.2012or later revisions of the above listed documents approved by EASA | none | 11.09.2012 |
| 10040290 | JET AVIATION AG, ZURICH AIRPORT BRA | | | | B200, B200C, B200GT, B200CGT B300, B300C | Mini Quick Access Recorder (QAR)Installation: Single or Dual Avionics P/N 804-1205 miniQAR Mk III | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.CS-23The certificated noise and/ or emissions levels of the original product are unchan (...) | Master Document List MDL-102 Rev. B 12.06.2012Instructions for Continued Airworthiness ICA-102 Rev. N/C of 16.04.2012or later revisions of the above listed documents approved by EASA | None | 22.06.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--|----------------|----------------|----------------------|---|--|---|---|---|---------------|
| 10042873 | JET AVIATION AG, ZURICH AIRPORT BRA | | | | B200, B200C, B200GT, B200CGT B300, B300C | BARO-VNAV Temp. Comp. Activation & RF Legs Authorization/Activation of Temperature Compensation on already approved APV BARO-VNAV System and RF Legs Authorization (This functionality can be used in the initial and intermediate approach segments, the final phase of the missed approach, SIDs and STARs) | The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Master Document List MDL-101 Rev. BAFM Supplement: Barometric Temperature Compensation/Correction FMS-101-1 Rev. IRAFM Supplement: Authorization for Radius to Fix Legs (RF-Legs) FMS-101-2 Rev. IR later revisions of the above listed documents approved by EASA | S/N: same as EASA TCDS with Rockwell Collins Proline 21 Avionics | 21.12.2012 |
| 10043147 | JET AVIATION BUSINESS JETS AG | EASA.A.S.03786 | REV.1 | | BOMBARDIER BD-700-1A10 | Activation of the Runway Awareness and Advisory System (RAAS) | | - Jet Aviation Declaration of Compliance ref BGLEX34461 dated 23/02/2009.- Jet Aviation Master Data List No. BGLEX34461-001, Revision IR, dated 23/02/2009.- Jet Aviation AFMS No. BGLEX34461-201-01, Revision IR. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The inst (...) | 25.02.2009 |
| 10015221 | JET AVIATION FLUGZEUGWARTUNG GmbH | EASA.A.S.03638 | REV. 1 | | 650 (CITATION III) | Primus 880 WX & MFRD installation/Removed existing Weather Radar Equipment- i.a.w. Installation Instructions, Doc. No. OPM-003 New Installed Weather Radar Equipment- Primus 880 Weather Radar System- Honeywell Multi Function Display, P/N 80-5204-XX-XI (...) | FAA TCDS A9NM The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged (...) | - Master Document List, Doc. No. MDL-059, Rev. B or latest approved Revision- AFMS, Doc. No. FMS-035, Rev. A or latest approved Revision- ICA, Doc. No. ICA-027, Rev. IR or latest approved Revision- Installation Instructions, Doc. No. OPM-003, Rev. B (...) | None | 23.02.2012 |
| 10015222 | JET AVIATION FLUGZEUGWARTUNG GmbH | EASA.A.S.03639 | REV. 2 | | 650 (CITATION III) | MK VII EGPWS Installation. Removed existing Equipment: i.a.w. Installation Instructions, Doc. No. OPM-004 New installed EGPWS Equipment: Honeywell MK-VII EGPWS i.a.w. Installation Instructions, Doc. No. OPM- 004 Revision 1: By revising the interface betwe (...) | FAA TCDS A9NM The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged (...) | - Master data List, Doc. No. MDL-060, Rev. B AFMS, Doc. No. FMS-036, Rev. A- ICA, Doc. No. ICA-028, Rev. IR- Installation Instructions, Doc. No. OPM-004, Rev. A or later revisions of the above listed documents approved by EASA | None | 23.02.2012 |
| 10015223 | JET AVIATION FLUGZEUGWARTUNG GmbH | EASA.A.S.03640 | REV. 1 | | 650 (CITATION III) | RAAS activation on the Honeywell MK-VII EGPWS i.a.w. Installation Instructions, Doc. No. OPM-005 Revision 1 has been issued to reflect the transfer of STC. | FAA TCDS A9NM The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged (...) | - Master Document List, Doc. No. MDL-061, Rev. A- AFMS, Doc. No. FMS-037, Rev. IR- ICA, Doc. No. ICA-029, Rev. IR- Installation Instructions, Doc. No. OPM-005, Rev. IR or later revisions of the above listed documents approved by EASA | None | 23.02.2012 |
| 10015224 | JET AVIATION FLUGZEUGWARTUNG GmbH | EASA.A.S.03641 | REV. 3 | | 650 | P-RNAV Compliance in installed GNS-XLS(S) FMS, i.a.w. Installation Instructions, Doc No. OPM-006 Revision 2 has been issued to reflect the removal of the limitation to C650 (Citation III up to SN-199), AFMS | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Jet Aviation Flight manual Supplement, FMS- 038, rev. B, date 12 December 2012.-Jet Aviation Master Document List, MDL-062, Rev. B, date 12 December 2012.-Jet Aviation Installation Procedure, OPM-006, Rev. A, date 12 December 2012.-Jet Aviation Maintenance (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 01.03.2013 |
| 10015225 | JET AVIATION FLUGZEUGWARTUNG GmbH | EASA.A.S.03642 | REV. 1 | | 650 (CITATION III) | FA2100 CVR installation/Removed Equipment- i.a.w. Installation Instructions, Doc. No. OPM-007 New Installed Equipment: FA2100 Cockpit Voice Recorder System i.a.w. Installation Instructions, Doc. No. OPM- 007 Revision 1 has been issued to reflect the transfer of STC. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Document List, Doc. No. MDL-063, Rev. A- AFMS, Doc. No. FMS-039, Rev. IR- ICA, Doc. No. ICA-031, Rev. IR- Installation Instructions, Doc. No. OPM-007, Rev. A or later revisions of the above listed documents approved by EASA | None | 23.02.2012 |
| 10040869 | JET AVIATION FLUGZEUGWARTUNG GmbH | | | | CESSNA 560XL | Installation of a Moving Map System. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Jet Aviation Master Data List: ref 2012-4-16, rev IR, date 30 July 2012.-Jet Aviation AFM Supplement: ref 2012-4-8, rev A, date 21 July 2012.-Jet Aviation Maintenance Manual Supplement, ref 2012-4-10, rev A, date 21 July 2012, or later revisions of the (...) | None. | 31.07.2012 |
| 10042792 | JET AVIATION FLUGZEUGWARTUNG GmbH | | | | 525 525A | WX-950 Stormscope installation. | CS 23 Amdt. 40 The Certification Basis for the original product remains applicable to this certificate/ approval. Derogation to latest amendment (Part 21A.101): Do not materially contribute to safety. The certified noise and/or emissions levels of th (...) | - Master Data List, Doc. No. 2016-6-16, Rev. IR, dated 12.12.2012.- AFMS, Doc. No. 2016- 6-8, Rev. A, dated 25.10.2012.- ICA, Doc. No. 2016-6-10, Rev. IR, dated 27.11.2012, or later revisions of the above listed documents approved by EASA. | None. | 18.12.2012 |
| 10044195 | JET AVIATION FLUGZEUGWARTUNG GmbH | | | | 550B | Installation of an Iridium Antenna on the top of the fuselage. According to Master Document List 2013-1-16, rev IR, dated 18 March 2013. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Jet Aviation Master Document List 2013-1- 16, rev IR, dated 18 March 2013.-Jet Aviation Maintenance Manual Supplement, 2013-1-10, Rev. A, dated 06 March 2013.-Jet Aviation Installation Procedure, 2013-1-1, rev IR, dated 25 January 2013, or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 25.03.2013 |
| 10045512 | JET AVIATION FLUGZEUGWARTUNG GmbH | | | | 525, 525A | P-RNAV Approval of UNS-1K | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Airplane Flight Manual Supplement 2013-6- 8, Rev. IR or later revisions of the above listed documents approved by EASA.-Engineering Order 2013-6, Rev. A, or later approved revisions-Installation Procedure 2013-6-1, Rev. IR, or later approved revisions | None | 01.07.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|-------------------|----------------|----------------------|--|---|--|--|--|---------------|
| 10014249 | JET AVIATION GMBH | EASA.A.S.01471 | | | LEARJET 55 LEARJET 55C | Installation of Flight Data Recorder F1000 | | -Engineering Order 2003-8, Rev B or later EASA approved revisions | | 01.08.2005 |
| 10014291 | JET AVIATION GMBH | EASA.A.S.01533 | | | IAI ASTRA SPX | Installation of Universal Unilink UL-601 | | Engineering Order, Report 2005-5 dated 16 February 2005, Rev. B or later EASA approved Revisions/Airplane Flight Manual Supplement, Report 2005-5-8, dated 09 March 2006 Rev. B or later EASA approved Revisions/Instruction for Continued Airworthiness, Maintenance Manual Supplement Rep. 2005-5-10 | Certification Basis:FAR Part 25, TGL Leaflet No. 15 and 16 | 19.04.2006 |
| 10014353 | JET AVIATION GMBH | EASA.A.S.01618 | | | CESSNA 560 (CITATION V) | Installation of Honeywell EGPWS Mark VIII | | Airplane Flight Manual Supplement Cessna C560 Citation V with Honeywell EGPWS Mark VIII, Revision A dated 23.02.2004 or later approved Revision, Jet Aviation Doc.No.2003-4-8/Aircraft Maintenance Manual Supplement Jet Aviation Doc. No. 2003-4-10, Rev.A | | 02.11.2005 |
| 10014393 | JET AVIATION GMBH | EASA.A.S.01683 | | | BEECH 1900 BEECH 1900C, AIRLINER BEECH 1900D | Installation of Universal UNS-1K+ | | AFM Supplement Report 2005-35-8, Rev. C or later EASA approved revision | Instructions for Continued Airworthiness, Doc. No. 2005-35-10 Rev. B | 31.08.2006 |
| 10014394 | JET AVIATION GMBH | EASA.A.S.01684 | | | BEECH 1900 BEECH 1900D | Installation of Universal TAWS and MFD 640 | | AFM Supplement Beechcraft 1900 and Series with Universal TAW/Sand MFD 640, Report 2005-36-8, Rev. B, or later approved revision | | 28.08.2006 |
| 10014401 | JET AVIATION GMBH | EASA.A.S.01696 | | | LEARJET 31 | Installation of Bluesky C-1000 SATCOM | | Jet Aviation Modification 2004-7 Revision A dated November 14, 2005 or alter approved revision | | 03.08.2006 |
| 10039198 | JET AVIATION ST. LOUIS, INC. | | | FAA STC ST01065WI-D | BD-700-1A10/11 | Installation of a Supplemental Crew Oxygen System | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | FAA STC No. ST01065WI-D dated 30 November 2011.Midcoast Master Data List 35-261G001 Revision K, dated 20 September 2010(Airplane Flight Manual Supplement(s) 35-261M001, Revision Original,dated 28 March 2002 (Config. 1); or 35-261M004, Revision NC, da (...) | None | 17.04.2012 |
| 10039199 | JET AVIATION ST. LOUIS, INC. | | | FAA STC ST02658CH-D | BD-700-1A10 | Installation of a Passenger Address System | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | FAA STC No. ST02658CH-D dated 30 November 2011.Jet AviationMaster Data List 35-411G001, Revision C, dated 5 December2008.Jet Aviation Airplane Flight Manual Supplement (AFMS) 35-411M001 RevisionNC, dated 6 December, 2008or later revisions ofthe above l (...) | None | 17.04.2012 |
| 10040607 | JET AVIATION ST. LOUIS, INC. | | | FAA STC ST01376WI-D | BD-700-1A10 | Installation of a Cabin Interior Arrangement, Passenger Oxygen Systemand Emergency Lighting | EASA.IM.A.009The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.EASA Generic CRIs D-01, D-02, D-03The requirements for environmental prot (...) | FAA STC No. ST01376WI-D dated 30 November 2011.Jet AviationIndex List 35-299G009, Revision N, dated 29 May 2011.Jet Aviation Master Drawing List 35-299G013, Revision D.Jet Aviation Airplane Flight Manual Supplement 35-174M003, Revision NC,dated 25 O (...) | Restricted to ASN 9267 (Configuration 13) | 13.07.2012 |
| 10041674 | JET AVIATION ST. LOUIS, INC. | | | FAA STC ST02868CH-D | CL-600-2B16 (CL-604 VARIANT) | Installation of Drain Mast | EASA.IM.A.023The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunch (...) | FAA STC No. ST02868CH-D.Jet Aviation St. Louis, Inc. MasterData List 41-260G001, Rev C dated 09February 2012.or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions ofthis decision) | None | 05.10.2012 |
| 10044411 | JET AVIATION ST. LOUIS, INC. | EASA.IM.A.S.02563 | REV. 1 | FAA STC ST02574CH-D | CL-600-2B16 | Installation of 40 Cubic-Foot Oxygen Bottle. | The Certification Basis (CB) for the original product remains applicabto this certificate/ approval.The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | FAA STC No. ST02574CH-D dated 30 January 2013.Jet Aviation St. Louis, Inc. Master Data List 41-261G001, Revision G,dated 30 November 2012.Airplane Flight Manual Supplement 41-261M001, Revision NC, dated 4 June2008.Instructions for Continued Airworth (...) | Restricted to ASN 5701 and subsequent.If this STC is installed in conjunction with the Cabin Interior STC No.EASA.IM.A.S.02539 (FAA STC No. ST02543CH-D), the Oxygen system AirplaneFlight Manual Supplement from the STC EASA.IM.A.S.02539 must be repla (...) | 15.04.2013 |
| 10046679 | JET AVIATION ST. LOUIS, INC. | | | FAA STC ST03229CH-D | BD-700-1A10, BD-700-1A11 | Cobham Aviator 700D SATCOM Upgrade | The Certification Basis (CB) for the original product remains applicabto this certificate/ approval.The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | FAA STC No. ST03229CH-D dated 18 June 2013.Jet Aviation St.Louis, Inc Master Data List 35-354G001 Rev A, dated 18June 2013.or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (orsubsequent (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noadverse effect upon the airworthiness ofthe produc (...) | 10.10.2013 |
| 10015882 | JET AVIATION ST. LOUIS, INC. | EASA.IM.A.S.02000 | | FAA STC ST01319WI-D | BOMBARDIER CL-600-2B16 (CL-604 | Addition of Runway Awareness & Advisory System-RAAS to Honeywell MK-V & MK-VII Enhanced Ground Proximity Warning System-EGPWS | | FAA STC ST01319WI-D dated April 22 2005.Midcoast Aviation Master Data List No. 34-327G003 Revision N/C dated April 22 2005 or later approved revision. | 1. Prerequisite: MK V or MK VII EGPWS must be installed at Software Level - 218-218 or greater.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previo (...) | 06.09.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|-------------------------|--|---|---|--|---|---------------|
| 10015883 | JET AVIATION ST. LOUIS, Inc. | EASA.IM.A.S.02001 | | FAA STC ST01318WI-D | BOMBARDIER CL-600-2B16 | Software upgrade to 218-218 of the Honeywell MK-V Enhanced Ground Proximity Warning System to support Runway Awareness & Advisory System (RAAS). | | FAA STC ST01318WI-D, dated 11 April 2005. Midcoast Master Data List No. 34-327G004 Revision N/C dated 11 April 2005 or later approved revision. | Prerequisite: MK V EGPWS must be installed per Bombardier Service Bulletin 604-34-020 or 604-34-025. | 06.09.2007 |
| 10016260 | JET AVIATION ST. LOUIS, Inc. | EASA.IM.A.S.02539 | | FAA STC ST02543CH-D | BOMBARDIER CL-600-2B16 | Installation of a Cabin Interior Arrangement, Passenger Oxygen System, and Emergency Lighting. | | Midcoast Aviation Inc. Master Data List 41-299G001, Revision H, dated April 9, 2008. Midcoast Aviation Inc. Airplane Flight Manual Supplements 41-261M002, Revision NC, dated April 09, 2008 and 41-424M001 Revision NC, dated April 9, 2008 or later approved revision. | Restricted to ASN 5701. | 03.06.2008 |
| 10016261 | JET AVIATION ST. LOUIS, Inc. | EASA.IM.A.S.02540 | | FAA STC ST4837WI-T | BOMBARDIER CL-600-2B16 | Installation of an Audio International Articulated Mounting Arm and Data Connectivity for an Electronic Flight Bag | | FAA STC No. ST02544CH-D dated 07 March 2008. Midcoast Aviation, Inc. Master Data List 41-343G001 Revision C, dated March 07, 2008, or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 03.10.2008 |
| 10016411 | JET AVIATION ST. LOUIS, Inc. | EASA.IM.A.S.02743 | | FAA STC ST00754WI | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 | Installation of Emergency Lighting System. | | Midcoast Aviation Certification Drawing List 34-415-G001 Revision E dated March 20, 1999 or later approved revisions. | This STC must be installed with Cabin interior arrangement per EASA STC No. EASA.IM.A.S.02745 and an oxygen system per EASA STC EASA.IM.A.S.02744 or other approved interior arrangement and oxygen system. | 20.03.2008 |
| 10016412 | JET AVIATION ST. LOUIS, Inc. | EASA.IM.A.S.02744 | | FAA STC ST00755WI | BOMBARDIER CL-600-2B16 | Installation of Passenger Oxygen System. | | Midcoast Aviation Certification Drawing List 34-261G001 Revision B dated March 15, 1999, or | This STC must be installed with Cabin interior arrangement per EASA STC No. EASA.IM.A.S.02745 and an emergency lighting per EASA STC No. IM.A.S.02743 or other approved interior arrangement and emergency lighting system. | 20.03.2008 |
| 10016413 | JET AVIATION ST. LOUIS, Inc. | EASA.IM.A.S.02745 | | FAA STC ST00756WI | BOMBARDIER CL-600-2B16 (601-3A) BOMBARDIER CL-600-2B16 (601-3R) BOMBARDIER CL-600-2B16 (604) | Cabin Interior Arrangement. | | Midcoast Aviation Certification Drawing List 34-299G001, Revision D, dated 18 March 1999, or later approved revision. | This STC must be installed concurrent with an emergency lighting system per EASA STC No. EASA.IM.A.S.02743 and an oxygen system per EASA.IM.A.S.02744 or other approved emergency lighting and oxygen system. | 20.03.2008 |
| 10016414 | JET AVIATION ST. LOUIS, Inc. | EASA.IM.A.S.02746 | | FAA STC ST00812WI | BOMBARDIER CL-600-2B16 (601-3A) BOMBARDIER CL-600-2B16 (601-3R) BOMBARDIER CL-600-2B16 (604) | Installation of the Allied Signal MK-V Enhanced Ground Proximity Warning System (EGPWS). | | Midcoast Certification Drawing List No. 34-327G001 Revision D, dated August 24, 1999. Midcoast Airplane Flight Manual Supplement, Document No. 34-327M001, Revision "Original", dated August 24, 1999, or later Approved Revisions to (1) or (2).< | Gamma Support Services Modification No. GSSL-MIN-CL604-226 (Including Gamma Support Services Flight Manual Change Sheet No. GSS-FMS-CL604-226 EASA approval number EASA.A.A.01831), or equivalent EASA approved modification, is a prerequisite for the EASA approval of this STC and must be incorporated. | 03.06.2008 |
| 10040724 | JET AVIATION ST. LOUIS, Inc. | | | FAA STC NO. ST02744CH-D | BD-700-1A10 | Installation of a Rockwell Collins Wireless Local Area Network (WLAN). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. ST02744CH-D dated 30 November 2011. Midcoast Aviation Index List 35-409G007, Revision NC dated 05 August 2009. Midcoast Airplane Flight Manual Supplement 35-409M004, Revision NC dated 06 August 2009 or later revisions of the above listed doc (...) | None. | 24.07.2012 |
| 10042519 | JET AVIATION ST. LOUIS, Inc. | | | FAA STC SA02789CE | MYSTERE FALCON MF 50 | EASA validation of FAA originally approved STC SA02789CE "Installation of a Bendix/King CAS-81 TCAS System". | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | See original FAA approved STC SA02789CE documents: "Jet Aviation Certification Drawing List No. AD910011, Revision F dated March 11th, 2011 and "Jet Aviation Report No 13-328M001, Revision F, FAA Flight Manual Supplement dated May 13th, 2011 and "J (...) | Aeroplanes modified in accordance with this STC must be maintained in accordance with Approved Instructions for Continued Airworthiness, Jet Aviation Report No. 13-027EG54, Section O - Airworthiness Limitations, Revision NC dated June 03rd 2011 or later (...) | 06.12.2012 |
| 10014336 | JET AVIONICS AVIATION & MARINE | EASA.A.S.01591 | | | CESSNA 650 CITATION III | BDI 004-05 Updating to TCAS II & ELT system and installation SSFDR | | Master Drawing List No. BDI 004-05-01 dated 18/04/2005. Technical Report No. BDI 004-05-02 dated 18/04/2005. AFMS for FDR No. BDI 004-05-10 dated 19/04/2005. AFMS S for TCAS II CAS 67ANo. BDI 004-05-15 dated 19/04/2005 | Applicable only if the STC No ST502CH of 16 may 1996 was previously installed | 11.10.2005 |
| 10014349 | JET AVIONICS AVIATION & MARINE | EASA.A.S.01614 | | | BEECH 58, BARON | Improving of avionics systems for updating FM immunity 8.33kHz channelling, B-RNAV, Mode S Transponder, TAWS, Weather Radar, Traffic Advisory System, ELT and entertainment system. | | Master Drawing List BDI 005-05-01 Rev. 0, dated 09/07/2007. | FM Supplement Sandel SN-3400, dated 09/07/2007. FM Supplement Sandel ST-3400, dated 09/07/2007. FM Supplement Garmin GTX 330, dated 09/07/2007. FM Supplement Garmin GNS 430, dated 09/07/2007. FM Supplement Garmin MX-20, dated 09/07/2007. FM Supplement L3 (...) | 08.08.2007 |
| 10014674 | JET AVIONICS AVIATION & MARINE | EASA.A.S.02834 | | | BOMBARDIER CL-215-6B11 | Installation of Radar Altimeter AA-300 System | | Master Drawing List BDI 002-06-01AFMS BDI 002-06-14 | N/A | 01.03.2007 |
| 10015044 | JET AVIONICS AVIATION & MARINE | EASA.A.S.03360 | | | CESSNA 500, CITATION | BDI 004-07 Installation of Mode S Transponder - Elementary Surveillance. | | BDI 004-07-01 Master Drawing List BDI 004-07-01-11 AFMS BDI 004-07-18 MMEL | | 05.03.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|----------------------|--|---|--|---|--|---------------|
| 10015237 | JET AVIONICS AVIATION & MARINE | EASA.A.S.03664 | | | CESSNA CITATION 501 | Installation of Mode S Transponder. | | According to Jet Avionics Master Document List Doc. No. BDI005-08-01 and Jet Avionics Installation Bulletin Doc. No. BDI 005-08-03, original issue, or later Revisions. To be operated in accordance with Jet Avionics AFM-S Document No. BDI 005-08-16, o (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 23.12.2008 |
| 10040231 | JET AVIONICS AVIATION & MARINE | | | | PA-31T CHEYENNE II | General avionic improvement/Installation - Replacement of GPS, MovingMap, Stormscope, Weather Radar system, TCAl I | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.CAR 3 and CS 23.365, 561, 571, 573, 867, 1309, 1311, 1351, 1431, 1529initial amendment. (...) | Master Drawing List BDI 001-10-01 Issue 19.03.2010, Maintenance Manual Supplement BDI 001-10-14 Issue 11.04.2011, Flight Supplement for GPS 155BDI 001-10-16/1 Issue 19 March 2010, Flight Manual Supplement for Sky497 BDI 001-10-16/2 Issue 19 March 2010 (...) | None. | 19.06.2012 |
| 10045880 | JET AVIONICS AVIATION & MARINE | | | | PA-31T CHEYENNE II | General Avionics Improvements/Installation/Replacement of COMM/NAV/GPS,TAWS/RMI, EHSI | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:The following paragraph(s) at a later amendment: CS23.561, 571, 573,867, 1309, 1311, 1321, 1322, 1431, 1529 at their initi (...) | Master Drawing List BDI 002-10-01 Ed. 1, Rev. 2 16.07.2013Maintenance Manual Supplement (ICA-ALS) BDI 002-10-14 Rev. 1 17.04.2013FMS for GNS-530, Doc. BDI 002-10-16/1 Rev. 0FMS for ST-3400, Doc. BDI 002-10-16/2 Rev. 1FMS SN 3308, Doc. BDI 002-10-16/1 (...) | Applicable only to S/N 8020038 | 25.07.2013 |
| 10045119 | JET AVIONICS S.R.L. | | | | PA-31T CHEYENNE II | General avionics Improvements/Installation/Replacement of COMM/NAV/GPOS,TAWS/RMI, DME, Audio Panel, Multifunctional Display | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:CS23.365, 561, 573, 867, 1309, 1311, 1351, 1431, 1529 at their initial amendmentThe requirements for environmental protect (...) | Master Drawing List BDI 003-10 Rev. 1 15.03.2013-Maintenance Manual Supplement BDI 003-10-14 Rev. 2 11.03.2013-Flight Manual Supplement GNS-530, ST3400, Flightmax FSD BDI 003-16-1(Rev. 0 30.11.2012), -2(Rev. 1 04.02.2013), -3 (Rev. 0 30.11.2012)-MM (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 31.05.2013 |
| 10016135 | JET ENGINEERING LLC | EASA.IM.A.S.02331 | | FAA STC ST01943LA | DC-9-81 (MD-81) DC-9-81 (MD-82) DC-9-81 (MD-83) DC-9-81 (MD-87) | Installation of a JET Engineering, Ltd. Advanced Jet Nozzles on the above airplane model(s) when powered by Pratt & Whitney JT8D-21/JT8D-217, 7A, -217C, 219 engines | | FAA STC ST01943LA amended 2/13/07/JET Engineering Master Drawing List (MDL) No. 201-1000 Rev. A, Airplane Flight Manual Supplement No. 200-495, Rev. A, Complete Data List, JET-CDL-201, Rev. B, Advanced JET Nozzle Installation, JET-205-310, Rev. NC, Weight (...) | This installation is done concurrently with STC EASA.IM.E.S.01012, installation of Jet Engineering, Ltd. Turbine Exhaust Cone Mixer (TECM), which modifies the Pratt & Whitney JT8D-217, JT8D-217A, -217C, -219 engines. | 14.06.2007 |
| 10016698 | JET ENGINEERING LLC | EASA.IM.E.S.01012 | | FAA STC ST01944LA | PRATT & WHITNEY JT8D-217 PRATT & WHITNEY JT8D-217A PRATT & WHITNEY JT8D-217C PRATT & WHITNEY JT8D-219 | Installation of a JET Engineering, Ltd. Turbine Exhaust Cone Mixer (TECM) | | - JET Engineering Master Drawing List (MDL) No. 205-1000, Rev. D, dated January 17, 2007, or later approved revisions - JET-TR-805 "Supplemental Structural Inspection Requirements", Revision NC, dated April 27, 2004, or later approved revisions | This installation must be done concurrently with STC EASA.IM.A.S.02331, installation of Jet Engineering, Ltd. Advanced Jet Nozzles which modifies the airplane engine installation. Approval of this change in type design applies to the above engine mod (...) | 05.03.2007 |
| 10042563 | JET ENGINEERING LLC | EASA.IM.A.S.01042 | | FAA STC ST01542LA | DC-9-81/-82/-83/-87, MD-88 | FAA STC No. ST01542LA Installation of JET Engineering, LLC Jet nozzles on the applicable models when powered by Pratt & Whitney JT8D-217/-219 engines. | | JET Engineering Master Drawing List (MDL) No 200-1000; Supplemental Maintenance Manual for the Jet Engineering Jet Nozzles installed on DC-9-80 series aircraft, JET-TR-401; Supplemental Structural Repair Manual to the McDonnell Douglas DC-9-80 series air (...) | 1. The installation should not be incorporated in any airplane unless its determined that the interrelationship between this installation and any previously approved configuration will not introduce any adverse effect upon the airworthiness of the airp (...) | 06.09.2005 |
| 10016549 | JETAIRE AEROSPACE, LLC | EASA.IM.A.S.02929 | REV. 2 | FAA STC ST01466LA-D | B737-300/-400 | Installation of fire protection systems, lower cargo compartment Class Dto Class C conversion. Original FAA STC ST01466LA-D for B737-300; this revision is implementing the validation of FAA STC ST00749LA-D for B737-400. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | B737-300- Hollingshead International Master Data List No. 2708-01, Revision NC, dated 26 JAN 2004- Hollingshead International FAA Approved Flight Manual Supplement No.2708-AFM, Revision NC, dated 26 JAN 2004- Hollingshead International Instructions for (...) | None. | 29.08.2012 |
| 10015707 | JETPROP, LLC. | EASA.IM.A.S.01698 | | LBA STC SA1189 | PIPER PA-46-310P PIPER PA-46-350P | Installation of the following engines, propellers and associated hardware: Pratt & Whitney PT6A-34 engine limited to 560 SHP at 2200 RPM Pratt & Whitney PT6A-35 engine limited to 560 SHP at 2190 RPM Pratt & Whitney PT6A-21 engine limited to 550 SHP at 2 (...) | | - Airplane Flight Manual Supplement for PA-46-310/350P (JetProp) Rev 7 or later approved revision- JetProp Maintenance Manual Doc No 560.1001, Change 2 or later approved revision- The instructions for the type are applicable as well | 1. This STC is valid only in connection with FAA STC SA00859AT (EMZ 0721/1076)2. This airplane/engine/propeller combination is life limited. The life limit for the modified airplane shall be determined by using the minimum value calculated by the foll (...) | 26.10.2005 |
| 10016000 | JETPROP, LLC. | EASA.IM.A.S.02158 | | ST00541SE | PIPER PA-46-310P PIPER PA-46-350P | Installation of PWC PT6A-34/-35/-21 turboprop engines in any combination with Hartzell Propellers: HC-E4N-3/E8501B-3, 5HC-E4N-3M/E8190KHC-E4N-3N/D8292HC-E4N-3N/D8292(B,K)-2 and MT-Propeller: MTV-16-1-E-C-F-R(P)/CFR206-58 on Piper PA-46-310P and PA-46-350P airplanes | | FAA Approved Airplane Flight Manual Supplement, change 20 or later approved change, Document No. 560.1002EASA, including EASA approval sheet, Approval No. EASA.IM.A.S.02158 Jetprop DLX Maintenance Manual, Document No. 560.1001, Change 8 or later Rocket (...) | PT6A-34, rated at 560 HP, 2200 RPM PT6A-35, rated at 560 HP, 2190 RPM PT6A-21, rated at 550 HP, 2200 RPM This airplane/engine/propeller combination is life limited. The life limit for the modified airplane shall be determined by using the minimum value (...) | 24.04.2008 |
| 10015892 | JOE'S AIRCRAFT MODS L.L.C. | EASA.IM.A.S.02012 | | FAA STC SA564NE | 172E, 172F, 172G, 172H, 172I 172K, 172L, 172M, 172N, 172P | Installation of a Textron Lycoming Model O-360-A44, -A4M or -A4N engine and a Sensinech Model 76EM8514-0-80 fixed Pitch metal propeller. | | Installation in accordance with Alphin Aircraft Inc. Drawing List No. AA-89-1 Revision B, dated August 28, 1980. Operation in accordance with FAA Approved Supplemental Flight Manual No. 1 for the Models 172I, K and L, revised February 16, 1990. FAA Ap (...) | | 08.11.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|---------------------|----------------|----------------------|---|---|---|---|--|---------------|
| 10016157 | JOHN DEERE AGRI SERVICES, INC | EASA.IM.A.S.02361 | | FAA STC SA00622DE | CESSNA 150L/M, 172M/N/P/Q CESSNA 182P/Q/R, F150L/M, CESSNA F172M/N/P, CESSNA F182P/Q | Installation of John Deere Agri Services, Inc. JD3 Aerial Imaging System (externally mounted camera / internally mounted computer) for aerial surveying of agricultural fields. | | Master Drawing List, Doc. No. JD3MDL, Rev. C, dated 11-Jan-2007, or later approved revision/Installation Instructions as listed in the ATL, Doc. No. EASA.IM.A.S.02361-ATL, Rev. 0, dated 26-Apr-2007, or later EASA approved revision, Airplane Flight Manu (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 04.05.2007 |
| 10034129 | KAWAK AVIATION TECHNOLOGIES, INC. | | REV. 1 | FAA STC SR01561SE | KAMAN K-1200 | Title: Auxiliary Hydraulic System, K-Max K1200 Helicopter Model 36300, DWG 36300, REV B, 25th June 2005 Description: Auxiliary hydraulic power source provides hydraulic power for external implements used in fire fighting, logging, and agricultural operation. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Approval of Rotorcraft Flight Manual Supplement: FAA Approved Rotorcraft Flight Manual Supplement for the Kaman Model K-1200 Helicopter, KATH-FM-632, Revision A, FAA-approved on 17th May 2007 or later revisions of the above listed documents approved by (...) | None | 21.03.2011 |
| 10015811 | KEITH PRODUCTS LP | EASA.IM.A.S.01880 | | FAA STC SA10144SC | CESSNA 206H, T206H | Installation of an engine driven vapour cycle air-conditioning system | | Install in accordance with Keith Products L.P. Master Document List Report Number DL-26-0100, Revision C, dated June 02, 2003 or later approved revision. Operate in accordance with Keith Products L.P. Approved Flight Manual Supplement (Certification R (...)) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 10.04.2006 |
| 10015824 | KELOWNA FLIGHTCRAFT Ltd. | EASA.IM.A.S.01896 | | SA04-32 | BOMBARDIER CL-215-6B11 | TCCA STC SA04-32 issue 3SPZ-4500 Honeywell AFCS and Universal UNS 1K/1L FMS | | - MDL K 16-00002- AFMS K126-102-014.1A, K126-102-014.2A and K126-102-014.3A | | 17.07.2006 |
| 10028498 | KEYSTONE HELICOPTER CORPORATION | P-EASA.IM.R.S.01562 | REV. 1 | FAA STC SR01127NY-D | S-76A, S-76B, S-76C | Installation of a Keystone Medical Interior - See Conditions | The Certification Basis for the original product remains applicable to this certificate/ approval | Modification Definition Keytech Installation drawings List IDL00549, dated September 16th 2002. Flight Manual Supplements For the S76A, S76B or S76C - RFMS No. 00549 dated September 16th 2002 (>-<)-or later revision of this RFMS No° 00549 document appro (...) | As specified in the approved RFMS. | 03.02.2010 |
| 10028501 | KEYSTONE HELICOPTER CORPORATION | P-EASA.IM.R.S.01563 | | FAA STC SR00910NY | S-76A, S-76B, S-76C | Structural provisions for the installation of a Spectrolab SX5 Searchlight to the left or right side of the S76. | The Certification Basis for the original product remains applicable to this certificate/ approval | Modification Definition Keytech Installation drawings List IDL00242, Rev A, dated February 7th 2000. Flight Manual Supplements For the S76A - RFMS No. 00242 dated Nov 19 1998 For the S76B - RFMS No. 00242.1 dated Nov 19 1998 For the S76C - RFMS No. 00242. (...) | As specified in the approved RFMS. | 18.01.2010 |
| 10028503 | KEYSTONE HELICOPTER CORPORATION | P-EASA.IM.R.S.01564 | | FAA STC SR00642NY | S-76A, S-76B, S-76C | Installation of a Liquid Oxygen System for Medevac use | The Certification Basis for the original product remains applicable to this certificate/ approval | Modification Definition Keytech Installation drawings List IDL00114 dated April 3rd 1997. Flight Manual Supplements For the S76A, S76B or S76C - RFMS No. 00114 dated April 3rd 1997 or later revisions of these documents approved by EASA in accordance with (...) | This approval should not be incorporated in any aircraft of this specific model on which other approved modifications are incorporated, unless it is determined that the interrelationship between this change and any other previously incorporated appro (...) | 18.01.2010 |
| 10029641 | KEYSTONE HELICOPTER CORPORATION | | | FAA STC SR02588NY-D | S-76C | Installation of Goodrich Auxiliary Utility Hoist | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: As specified on FAA STC SR02588NY-D The certificated noise and/or emissions levels of t (...) | - RFMS FM01464 Rev 0 dated 2 Oct 2009 or later approved revision.- Instructions for Continued Airworthiness IC01464 Rev 0 dated 15 May 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (...) | As specified in the approved RFMS | 08.04.2010 |
| 10030668 | KEYSTONE HELICOPTER CORPORATION | | | FAA STC SR02597NY-D | S92A | Installation of a tail mounted, forward facing camera. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | IDL01702 revision - dated 12 March 2010 - Installation Drawing List: IC01702 rev - dated 1 January 2010 - Instructions for Continued Airworthiness: FM01702 rev - dated 21 May 2010 - Flight Manual Supplement. | None. | 30.06.2010 |
| 10043043 | KEYSTONE HELICOPTER CORPORATION | P-EASA.IM.R.S.01536 | | FAA STC SR02430NY-D | S-76C | Installation of a Honeywell Chadwick EVXP Vibration Monitoring System | | - Report Document List RDL01449 Rev A or later approved revision.- Instruction for Continued Airworthiness IC01449 Rev A.- Flight Manual Supplement FM01729 Rev- | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The (...) | 16.06.2009 |
| 10043051 | KEYSTONE HELICOPTER SERVICES | EASA.IM.R.S.01113 | | FAA STC SR01963NY-D | S76B/C | Installation of the WesCam I2DS200 Camera System | | Installation Drawing List IDL00896 rev" * dated May 10, 2005 or later approved revision/ Flight Manual Supplement FM00896 rev - dated May 24, 2005 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate sha (...) | 13.03.2006 |
| 10043052 | KEYSTONE HELICOPTER SERVICES | EASA.IM.R.S.01112 | | FAA STC SR01962NY-D | S76B/C | Installation of Rockwell Collins MDF-124F (V2) Direction Finder System | | Installation Drawing List IDL00895 rev" * dated May 10, 2005 or later approved revision/ Flight Manual Supplement FM00895 rev - dated May 24, 2005 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate sha (...) | 13.03.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|---------------------|----------------|------------------------|---|---|---|--|---|---------------|
| 10043053 | KEYSTONE HELICOPTER SERVICES | EASA.IM.R.S.01111 | | FAA STC SRO1961NY-D | S76C | Installation of a Vertical Visibility Door Window in the pilot's door | | Installation Drawing List IDL00929 rev A dated May 4, 2005 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate sha (...) | 13.03.2006 |
| 10043055 | KEYSTONE HELICOPTER SERVICES | EASA.IM.R.S.01110 | | FAA STC SRO1959NY-D | S76A/B/C | Installation of Grimes Search Lights | | Installation Drawing List IDL00928 rev A dated May 10, 2005 or later approved revision; Flight Manual Supplement FM00928 rev - dated May 24, 2005 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate sha (...) | 13.03.2006 |
| 10045868 | KLM ENGINEERING & MAINTENANCE | | | | 747-400 | KLM 747-400 cabin reconfiguration/Replacement of Business Class seats with new, full-flat Business Class seats + related changes to cabin furniture to accommodate the new cabin layout. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:- Special Condition(s): CRI D-GEN02 PTC - Application of heat release and smoke density requirements to seat materials- th (...) | EA 012613, Revision 2, 10 July 2013 - Certification Plan/CA-747-012613-001, Revision Org, 15 July 2013 - Instructions for Continued Airworthiness MDL-012613, Revision 2, 25 July 2013 - Master Data List CE-18552, Revision 2, 25 July 2013 - Compliance Ch (...) | EASA Project number 0010018346-001 (added for reference purposes for the prerequisite STC's) The following STC's should be installed simultaneously: STC EASA.IM.A.S.03182, Delta Engineering - Update of Panasonic IFE/IS/SS configuration (amendment of pr (...) | 25.07.2013 |
| 10041473 | KMLEX AVIATION INC. | | | TCCA STC Q-LSA10-047/D | FALCON 50EX | Relocation of Radio Altimeter Converter/Validation of TCCA STC N° Q-LSA10-047/D issue dated November 04, 2010. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List MDL30158002 - Relocation of Radio Altimeter Converter on Falcon 50 aircraft S/N 278 (>-<-) revision NC dated November 04, 2010. TCCA STC N° Q-LSA10-047/D issue dated November 04, 2010 or later revisions of the above listed documents (...) | Restricted to S/N 278 | 19.09.2012 |
| 10041475 | KMLEX AVIATION INC. | | | TCCA STC Q-LSA10-048/D | FALCON 50EX | Replacement of Alt lavatory faucet adapter - Dassault Falcon 50EX aircraft MSN 278/Validation of TCCA STC N° Q-LSA10-048/D issue dated November 04, 2010. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List MDL30159002 - Replacement of Alt lavatory faucet adapter on Falcon 50 aircraft S/N 278 (>-<-) revision NC dated November 04, 2010. TCCA STC N° Q-LSA10-048/D issue dated November 04, 2010 or later revisions of the above listed docum (...) | Restricted to S/N 278 | 19.09.2012 |
| 10015415 | KNH AVIATION SERVICES, INC. | EASA.IM.A.S.01136 | | FAA STC ST09993SC | FAIRCHILD DORNIER DO 328-300 | FAA STC ST09993SC; 12 Seats configuration (extended to 14 seats through minor change, part of Avrocraft CN-F 0535) Installation of interior shell, cabinetry, seats, card tables, monitors, entertainment system, system wiring, and lavatory in accordance (...) | | FCOM , Temp Rev No 05-11, effective for S/N 3199, AFMS Doc Nr AFMS-D03-011578, Revision NC, dated June 10, 2002 or later FAA approved revision | 1. This STC may be installed only in combination with Change Notice CN-F0535, F0040-F0125 owned by Avrocraft Germany and RUAG minor Changes #In Seat Power Supply and FI Switch# and #Installation of Double Latches on drawers# 2. The water pumpin (...) | 20.06.2006 |
| 10015482 | KNH AVIATION SERVICES, INC. | EASA.IM.A.S.01242 | | FAA STC ST09468AC | DORNIER D328-100 | Installation of a corporate/executive interior | | -Fairchild Aircraft Engineering Change Package (ECP) 3000, Revision AN dated 28/08/1999-FAA STC ST09468AC, reissued 18 April 2005 | 1. According to the application as forwarded by FAA letter dated 27 April 2005 and FAA STC ST09468AC, the approval is limited to the aircraft S/N 30952. Furthermore this installation has to be performed in combination with the installation of FAA STC (...) | 15.08.2005 |
| 10028304 | KNOTS 2U, LTD. | P-EASA.IM.A.S.03267 | | | PA-28-151, PA-28-161, PA-28-181, PA-28-201T, PA-28-236, PA-28R-201 PA-28-RT-201T, PA-32-301 PA-28R-201T, PA-28RT-201, PA-32-301 T, PA-32R-301, PA-32R-301T | Piper Wing Tips with Landing / Recognition Lights | CS-23.B NSC | Installation instructions as listed on AML-SA2356NM issued 2008-08-14 or later FAA approved revision. | None | 16.12.2009 |
| 10031230 | KNOTS 2U, LTD. | LBA EMZ SA1221 | | FAA SA1217GL | 1. PA-32-301FT, PA-32-301XTC 1. PA-32R-301(HP) 1. PA-32R-301(TC) 2. PA-32-301, PA-32-301T 2. PA-32R-301, PA-32R-301(SP) | Installation of wing root fairings in accordance with FAA STC SA1217GL (Revision of LBA EMZ 1221, issue 1, dated July 20, 2001). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Knots2U, Ltd. Installation and Maintenance Manual, document number 32WR-M, revision E, dated May 2, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 03.08.2010 |
| 10031235 | KNOTS 2U, LTD. | | | FAA STC SA1521GL | 1. PA-32-301FT, PA-32-301XTC 1. PA-32R-301(HP) 1. PA-32R-301(TC) 2. PA-32-301, PA-32-301T 2. PA-32R-301, PA-32R-301(SP) | Installation of flaps, aileron, flap/fuselage and stabilator gap seals in accordance with FAA STC SA1521GL | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Knots2U, Ltd. Installation and Maintenance Manual, document number 32FGS-M, revision F, dated May 2, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 03.08.2010 |
| 10031236 | KNOTS 2U, LTD. | LBA EMZ SA1222 | | FAA SA1195GL | 1. PA-32-301FT, PA-32-301XTC 1. PA-32R-301(HP) 1. PA-32R-301(TC) 2. PA-32-260, PA-32-300 2. PA-32-301, PA-32-301T 2. PA-32R-300, PA-32RT-300 2. PA-32R-301(SP), PA-32S-300 2. PA-32R-301T(SP), PA-32R-300 2. PA-32RT-300T | Installation of Flap Hinge Fairing seals in accordance with FAA STCSA1195GL (Revision of LBA EMZ 1222, issue 1, dated July 20, 2001). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Knots2U, Ltd. Installation and Maintenance Manual, document number 32HF-M, revision D, dated February 10, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 03.08.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|----------------|----------------|----------------------|-----------------------------|--|---------------------|--|--|---------------|
| 10014885 | KORFF LUFTFAHRT - RAINER KORFF | EASA.A.S.03113 | | | GROB G 109 GROB G 109 B | Changes:- at the luggage compartment to be used as basis for instruments, - additional cable routing in the wings- reinforced attachment points at the lower side of both wings for under wing pots- reinforcements of the root tips- operation with exter (...) | | # Modification Bulletin "Änderung am Stück: für Grob G 109B, S/N 6219" dated February 15th 2006, prepared by companyKorff Luftfahrt für the STC (Major Change).# Flughandbuch: Anhang II "Messflugzeug" (Flight Manual Annex II)# Wartungshandbuch: Seite 24 (Maintenance Manual: Page 24). | 1. Limited to S/N: 62192. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and #Associated Technical Documentation; Compatibility with other aircraft/eng (...) | 30.04.2007 |
| 10014470 | KUERZI AVIONICS AG | EASA.A.S.02531 | | | LEARJET 60 | Installation of SSFDR and MFDInstallation of a Honeywell Solide State Flight Data Recorder and Teledyne Mini Data Acquisition Unit in Learjet Model 60 | | Master Document List No. MDL-200514764. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 11.12.2006 |
| 10014482 | KUERZI AVIONICS AG | EASA.A.S.02545 | | | CESSNA 340A | Avionic Retrofit | | Master Document/Drawing List, Doc No. MDL-200514568 Rev. B, dated 18.04.2006AFMS, Doc No. FMS-200514568, Original, dated 06.03.2006 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 25.04.2006 |
| 10014569 | KUERZI AVIONICS AG | EASA.A.S.02673 | | | BOMBARDIER DHC-6 SERIES 300 | Installation of KMH-880, GPS-400 & KMD-850. | | MDL-200615247 Rev. C or later approved revisions. | Limited to S/N 0841. | 13.04.2007 |
| 10014716 | KUERZI AVIONICS AG | EASA.A.S.02892 | | | PILATUS PC-12/45 | Honeywell GNS-XLS FMS InstallationBy installing a single GNS-XLS Flight Management System with GPS and DME sensor. | | Master Document List, Doc. No. MDL-200616263, Rev. N/CAAFMS,Doc. No. FMS-200616263, Rev. N/Instructions for Continued Airworthiness, Doc. No. ICA-200616263, Rev. N/C | HB-FPI S/N 551. Valid for this S/N only. | 23.04.2007 |
| 10014806 | KUERZI AVIONICS AG | EASA.A.S.03010 | | | PILATUS PC-12/47 | HF 1050 & AFFS Installation by installing a new HF-1050 HF COM System with a PS440 HF Control Panel and a new FSK AFFS (Automated Flight Following System). | | Declaration of Compliance, Doc. No. DOC-200616236, Master Document List, Doc. No. MDL-200616236-AFMS for the HF-1050, Doc. No. FMS-200616236.1 AFMS for the FSK AFFS, Doc. No. FMS-200616236. | S/N 795 for initial installation.Approval valid for S/N 795and any other PC-12/47. | 22.03.2007 |
| 10014878 | KUERZI AVIONICS AG | EASA.A.S.03105 | | | PIPER PA-31T | Installation of Dual Mode-S XPDR & TASIInstallation of- Dual Mode S Transponder GTX-330D with diversity antenna.- Traffic Advisory System SKY497 | | Master Document List, Doc No. MDL-200716854 | Reg. HB-LNLS/N: 31T-8020083Valid for this S/N only | 11.05.2007 |
| 10014945 | KUERZI AVIONICS AG | EASA.A.S.03205 | | | CESSNA 340A | Installation of a Traffic Advisory System:By installing a SKY497 Skywatch Traffic Advisory System with appropriate antenna. Traffic information will be displayed on existingsystems. | | Master Document List, Doc. No. MDL-200717371, Declaration of Compliance, Doc. No. DOC-200717371, AFMS, Doc. No. FMS-200717371, Rev. N/C or latest approved revision. | | 28.08.2007 |
| 10014974 | KUERZI AVIONICS AG | EASA.A.S.03260 | | | PILATUS PC-12/47 | FMS and DME installation.By installing a new GNS-XLS FMSa new DM-441B DME | | Master Document List, Doc. No. MDL-200717475, Rev. N/C or latest approved Revision.AFMS, Doc. No. FMS-200717475, Rev. N/C or latest approved Revision.Instruction for continued Airworthiness, Doc No. ICA-200717475, Rev. N/C or latest approved Revision. | | 26.11.2007 |
| 10015230 | KUERZI AVIONICS AG | EASA.A.S.03648 | | | PILATUS PC-12/47E | FDR and CVR Installation. | | 1. MDL-200830253 Rev. A or later approved revision2. FMS-200830253 Rev. N/C or later approved revision3. ICA-200830253 Rev. A or later approved revision | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 31.10.2008 |
| 10015275 | KUERZI AVIONICS AG | EASA.A.S.03770 | | | PILATUS PC-12/45 | GPS 400W and GMA 347 InstallationRemoved Equipment:- KLM 90B GPS- Existing GPS Antenna- KMA 24H Audio PanelNew Installed Equipment- GPS 400W GPS- GA 35 GPS/WAAS Antenna-GMA 347 Audio Panel | | - Master Document List, Doc. No. MDL-200830501, Rev. N/C or latest approved Revision- Declaration of Compliance, Doc. No. DOC-200830501, Rev. N/C- AFMS, Doc. No. FMS-200830501, Rev. N/C or latest approved Revision- ICA, Doc. No. ICA-200830501, Rev. N/C or latest approved Revision | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 02.03.2009 |
| 10015277 | KUERZI AVIONICS AG | EASA.A.S.03789 | | | PIPER PA-23-250 | Dual GNS430W, GMA340, KN 63 InstallationOut: Dual KX-155 COM/NAV systems, Trimble-2000 approach GPS system, KMA-24H Audio system, MKR-101 marker system, KN-65 DME systemIn: Dual GNS-430W system, GMA-340 Audio/MKR system, KN-63 DME system. | | 2. Master document list MDL-200830709 Rev. N/C or later approved revisions, AFM supplement FMS-200830709 Rev. N/C or later approved revisions,2. Instructions for Continued Airworthiness ICA-200830709 Rev. N/C or later approved revisions | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 08.04.2009 |
| 10017221 | KUERZI AVIONICS AG | EASA.R.S.01382 | | | BELL 206B | Hellas installation:As a convertible EADS HELLAS (Helicopter Laser Radar) System that provides the following functions:-Raw data acquisition-System development-System demonstrationThe rotorcraft can be operated either in a "Normal" or a "HELLAS" configuration. | | Master Document List, Doc. No. MDL-200717081, Rev. A or latest approved revision,Flight Manual Supplement, Doc. No. FMS-200717081, Original issue or latest approved revision. | Reg.: D-HEDB S/N: 4197Valid for this S/N only | 25.10.2007 |
| 10017239 | KUERZI AVIONICS AG | EASA.R.S.01425 | | | EUROCOPTER FRANCE SA 315B | NVIS Conversion.Includes Rad. Altimeter and Mode S ELS Transponder installation. | | Master Document List MDL-200717179 Rev. B or later approvedrevision,Flight Manual Supplement FMS-200717179 Rev. B or later approved revisions. | Reg.: HB-XII S/N: 2551Mode S Transponder system alleviationof flight status according JAA TGL-13 note 10 granted. | 06.08.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------|------------------|----------------|----------------------|----------------------|---|---|---|---|---------------|
| 10017262 | KUERZI AVIONICS AG | EASA.R.S.01471 | | | BELL 206B | Radar altimeter installation including adapter plate. | | # FMS-200718268 original issue dated 17 December 2008 or later EASA approved revision # Flight Manual Supplement: # ICA-200718268 revision N/C dated 28 November 2008 or later EASA approved revision # Instructions for Continued Airworthiness: # MDL-200718268 revision N/C dated 1 December 2008 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 18.12.2008 |
| 10027877 | KUERZI AVIONICS AG | P-EASA.A.S.03965 | | | LEARJET 55 | Installation of Satellite Tracking System SAT-111 | As specified in PCP-200931957, rev. A (or later approved revisions) The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged (...) | As specified in MDL-200931957, rev. N/Cor later revisions of the above listed documents approved by EASA | None | 11.11.2009 |
| 10033212 | KUERZI AVIONICS AG | | REV. 1 | | MDD 500N | MD520N Gyro camera & equipment inst. (Gyrocam "Turret Assy" for filming or live broadcasting) | CAR 6 dated December 20, 1956, including Amendments 6-1 through 6-4 and Special Conditions. The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged (...) | - Flight Manual Supplement Doc. No. FMS-201032936; or later revisions of the above listed documents approved by EASA. - Operating Instructions Doc. No. OPI-201032936; - Master data list Doc. MDL-201032936; | Reg.: \\\ S/N: LN052; LN 068; LN069 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect (...) | 14.02.2013 |
| 10033622 | KUERZI AVIONICS AG | | | | AS 350 B2, AS 350 B3 | G500H Avionic Retrofit (Option 1 & 2) Option 1: Original OEM installed - Airspeed Indicator - Gyro Horizon - Altimeter - Turn & Bank Indicator - Directional Gyro - Vertical Speed Indicator - Garmin G500H system including PFD/MFD GDU 620 - Attitude heading Refer (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI A-01 - G500H Avionics Retrofit, Issue 3, 11 January 2011. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement Doc. No.: FMS-201034169 Rev. Original or later revisions of the above listed documents approved by EASA. Master Document List Doc. No.: MDL-201034169 Rev. N/Instructions for Continued Airworthiness Doc. No.: ICA-201034169 Rev. N/C | None | 31.01.2011 |
| 10034392 | KUERZI AVIONICS AG | | | | MYSTÈRE-FALCON 50 | FMS upgrade/Update of the existing UNS-1k to UNS-1LV. Installation of a second FMS. Upgrade of the existing Collins VIR-31 1/2 and DME-41 1/2 to VIR-32 and DME-42 so that ARINC429 can be used to interface with the dual FMS. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document/ Drawing List doc N° MDL-201034519 Rev A dated 30/03/2011 Master Package List doc N° MPL-201034519 Rev N/C dated 30/03/2011 Aircraft Flight Manual Supplement doc N° FMS-201034519 Rev 0 dated 28 March 2011. Instructions for Continued Airwo (...) | Restricted to MF50 S/N 183. | 31.03.2011 |
| 10035249 | KUERZI AVIONICS AG | | | | PC-6-B2-H4 | Cold weather operation package (230VAC or 110VAC) By installation of: - One TANIS TAS200 Battery Heater system - One TANSTAS300 Turbine Preheat system - One Power receptacle in the cabin front right | CAR 3, Amdt: 3-1 through 3-5 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS or ESFs: For this change, voluntary compliance to CS23 Am (...) | Master Document List: MDL-201110187, dated 08.06.2011 Project Installation Manual: PIM-201110187, dated 08.06.2011 Instruction for continued airworthiness: ICA-201110187, dated 08.06.2011 Airplane Flight Manual Supplement: FMS-201110187, dated 08.06.20 (...) | S/N: all | 08.06.2011 |
| 10035488 | KUERZI AVIONICS AG | | | | DHC-6-300 | Installation of two Aspen Electronic Flight Display (EFD1000 Pro, one Garmin GNS530W (GPS, COM, NAV), one Garmin GNS430W (GPS, COM, NAV), one Garmin GTX330 (XPDR) and one Garmin GTS800 (TAS) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. FAR-23, Amdt: Delect to comply FAR 23.1308 Amdt. 57 FAR 23.1309 Amdt. 41 FAR 23.1311 Amdt. 41 (...) | Master document list MDL-201034120, Initial issue Product Installation Manual PIM-201034120, Initial issue Flight Manual Supplement FMS-201034120, Initial issue Instruction of Continued Airworthiness ICA-201034120, Initial issue for later revisions of the above listed documents approved by EASA | None | 27.06.2011 |
| 10036481 | KUERZI AVIONICS AG | | | | MF50 | P-RNAV and RNP10 certification/Certification of P-RNAV and RNP10 for a Falcon 50 based on existing EASASTC 10034392. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List doc N° MDL-201110379 Rev N/C dated 09 September 2011. Aircraft Flight Manual Supplement doc N° FMS-201034519 Rev 1 dated 21 June 2011, or later revisions of the above listed documents approved by EASA | EASA STC 10034392 is a required pre-requisite. | 12.09.2011 |
| 10036940 | KUERZI AVIONICS AG | | | | EC 130 B4 | G500H Avionic Retrofit/Out: Original OEM installed Horizon and Directional Gyro system (option 1 and 2) Original OEM installed Airspeed Indicator (only option 1) and Altimeter (only option 1). In: Garmin G500H system including PFD/MFD GDU 620, Attitude H (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 27 Amdt 0 for the change. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement Doc. No.: FMS-201110617 Rev. Original or later revisions of the above listed documents approved by EASA. Master Document List Doc. No.: MDL-201110617 Rev. N/Instructions for Continued Airworthiness Doc. No.: ICA-201110617 Rev. A. | S/N: all; minimum one G500H compatible GPS/WAAS navigator is a prerequisite. | 03.11.2011 |
| 10039239 | KUERZI AVIONICS AG | | | | MF900 | SATCOM Upgrade to WiFi/Activation of the WLAN option of the existing AVIATOR 300 SATCOM system on MF900 S/N 118 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List doc No. MDL-201110583 Rev N/C dated 16 April 2012. Aircraft Flight Manual Supplement doc No. FMS-201110583 Rev 0 dated 29 September 2011, or later revisions of the above listed documents approved by EASA | Limited to MF 900 S/N 118 | 19.04.2012 |
| 10040176 | KUERZI AVIONICS AG | | | | B737-300 | GSM mail tracker installation: a small GSM device integrated into the aircraft and able to report its current position by means of GSM cell information. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. - CRI F-01 (Special Condition/Interpretive Material) Issue 03 "Lithium batteries instal (...) | Project and Order Description POD-201033911 rev. B Master Document List 201033911 rev. N/Instructions for Continued Airworthiness ICA201033911 rev. N/Cor later revisions of the above listed documents approved by EASA | None | 18.06.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|---------------------------|--|---|---|--|---------------|
| 10041972 | KUERZI AVIONICS AG | | | | R.90-230RG | Avionic upgradeRemoved:- Original flight instrument including KCS55 pictorial navigation system, attitude indication (with light director), electrical attitude indication, encoding altimeter, 2x mechanical AS, VS and turn and bank. - Original vacuum sy (...) | FAR 23., Amdt: 35The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.SCs or ESFs: For this change compliance to FAR 23 Amdt 35 and FAR 23. (...) | Master Document List: MDL-201110675, Rev. B, dated 26.10.2012;Instructions for continued airworthiness: ICA-201110675, Rev. N/C, dated 27.08.2012;Airplane Flight Manual Supplements for - Garmin 500 PFD/MFD: FMS-201110675.1 Rev. 0, dated 18.09.2012 - G (...) | None. | 29.10.2012 |
| 10042467 | KUERZI AVIONICS AG | | | | PC-12/45 | Replacement of obsolete systemsReplacement of:- Bendix/King EFIS40/50 Systems with Universal EF1890R 3-Display Configuration (PFD/ND/CPFD) incl. required controls.- Universal UNS-1K FMS with UNS-1Lw FMS incl. LPV Monitor as primary NAVSource Installa (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: For this change compliance to FAR 23 Amdt 42 and FAR 23.562 (d) Amdt 44, FAR23.571(c) A (...) | Master Document List: MDL-201110853, Rev. N/C, dated 30.11.2012;Instructions for continued airworthiness: ICA-201110853, Rev. A, dated 26.11.2012;Airplane Flight Manual Supplement: FMS-201110853, Rev. N/C, dated 30.11.2012;or later revisions of the above listed documents approved by EASA. | S/N: 134 ; HB-FOG | 03.12.2012 |
| 10043605 | KUERZI AVIONICS AG | | | | MD 500N | Avionic upgrade with Garmin 500HRemoved equipment:- OEM Attitude indicator(->,-<)-, OEM Directional Gyro(->,-<)-, OEM Vertical Speed indicatorInstalled equipment:-, Garmin 500H System which includes:-, GDU 620 PFD/MFD,-, GRS 77H AHS,-, G (...) | The Certification Basis for the original product and the following voluntary compliance to FAR 27 was shown.The actual Amdt. used for each § is indicated after the FAR §. E.g.FAR27.1317 /42 means Amdt 42 was used.FAR27.29 /0, FAR27.143 /21,FAR27.251 (...) | Flight Manual Supplement: FMS-200932756, Original Issue, dated 11.02.2013;or later revisions of the above listed documents approved by EASA. - Master Data List: MDL-200932756, Rev. A, dated 11.02.2013; - Instructions for continued airworthiness: ICA-200932756, Rev. A, dated 11.02.2013. | S/N: all- At least one SBAS capable Garmin GPS unit must be installed as prerequisite.- This STC can only be installed on helicopter with a left hand command "T"- Rotorcraft modified under this STC are restricted to VFR only, including rotorcraft that may | 14.02.2013 |
| 10044092 | KUERZI AVIONICS AG | EASA.R.S.01280 | | | SA 330 J | NVIS Cockpit | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement Doc. No. FMS-200717408 Rev. N/Cor later revisions of the above listed documents approved by EASAMaster Document List Doc. No. MDL-200717408 Rev. CInstructions for Continued Airworthiness Doc. No. ICA-200717408 Rev. C | Refer to applicable RFM for list of equipment mandatory for Night Vision/IMG System certification.Installation of NVIS/NVG compatible lighting does not include or imply approval for flight operations with Night Vision Goggles (NVG). Theoperator must (...) | 18.03.2013 |
| 10044343 | KUERZI AVIONICS AG | | | | 60 | Skytrac ISAT-100 SatCom & flight following system installation: Skytrac ISAT-100 Iridium Transceiver/GPS Receiver; DVI-250 C control unit; CDP-250G display unit and Dual Iridium/GPS antenna | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List Doc. No.: MDL-200931669 Rev. N/C of 03.04.2013;Instr. for Cont Airworthiness ICA-200931669 Rev. N/CoF 03.04.2013;or later revisions of the above listed documents approved by EASA. | S/N: allPrior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 05.04.2013 |
| 10045344 | KUERZI AVIONICS AG | | | | FALCON 7X | GSM & Wi-Fi IntegrationNew installed equipment:- TriGNoSys, 21T810-00-01-00 AEROBOTS+ (GSM Base Transceiver Station,BTS)- TriGNoSys, 21T820-00-01-00 NCU (Network Control Unit)- VT Miltpoe, 902693-1 nMAP (Cabin Network Router)- Mobile Mark, RMM-UMB (...) | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List, Doc. No. MDL-201110599, Rev. A, 08.06.2013 - AFMS, Doc. No. FMS-201110599, Rev. 0, 28.05.2013 - ICA, Doc. No. ICA-201110599, Rev. N/C, 29.05.2013;or later revisions of the above listed documents approved by EASA. | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 19.06.2013 |
| 10017171 | KUERZI AVIONICS DEUTSCHLAND GMBH | EASA.R.S.01282 | REV. 2 | | EUROCOPTER FRANCE SA 330J | Avionics UpgradeGPS, Monitor, Cabin Intercom, EMS, SAT111 Tracking System, HF Radio, WX Radar, Power Connector, FM Radio | | Master Document and Drawing List No. MDL-200616059 Rev. C or later approved revisionsFlight Manual Supplement Garmin GPS165 No. FMS-200616059 Rev. A or later approved revisions | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 06.01.2009 |
| 10028707 | KUERZI AVIONICS DEUTSCHLAND GMBH | P-EASA.A.S.03752 | REV.2 | | PC-6-B2-H4 | Transfer of STC 10028707Garmin G950 Installation.Integration of a new Garmin G950 Integrated Flight Deck and legacy systems on the Pilatus PC-6/B2-H4 model.The G950 system (a derivative of the G1000 system) combines primary flight instrumentation, air/cr (...) | CS 23See CRI A-01The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List No. MDL-200830764 Rev. N/C and MDL-201033183 Rev.N/C for PRNAV/Flight Manual Supplement No. FMS-200830764 Rev. B;Instruction for Continued Airworthiness No. ICA-200830764 Rev. N/C dated 18.01.2010;or later revisions of the above listed documents approved by EASA | Required pre-modifications:- Pilatus PC-6 pitch trim warning system according SB-180-Electrical stall warning system, type Safe Flight 1-02-0005 No. 164(Transmitter) and C-73113 (Buzzer)- Electrical power distribution system (MCA 06/24/130)- Electr (...) | 30.03.2011 |
| 10038677 | L D AVIATION PRAGUE, s.r.o. | | | | F172M F172N F172P | Avionics upgrade - installation of:GMA 340 (audio), SL 40 (VHF COMM), SL 30 (VHF NAV/COMM),MD 200-306 (CDI indicator), GTX 328 (Mode S XPNDR), CI 102 (NKRAntenna), Kanned 406 AF Compact (ELT), AV 230 (ELT antenna), CI 105(XPNDR antenna). | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:FAR Part 23 Amdt 50, dated 10 March 1996 including special conditions:- SC-F23.1306 (...) | Cessna F172 M, N, P Flight Manual Supplement No. S1 LDA 02/11/C172GMA 340 Audio Panel, issued: May 30, 2011- Cessna F172 M, N, P Flight Manual Supplement No. S2 LDA 02/11/C172SL30 VHF COMM Transceiver/ NAV Receiver, issued: May 30, 2011- Cessna F1 (...) | None. | 07.03.2012 |
| 10015481 | L-3 COMMUNICATIONS / AVIATION | EASA.IM.A.S.01240 | | | B757-200 | Additional Parameters for Flight Recording Systems Ass.:Flight System Report 98-10-24 Revision C dated December 2004 or later approved revision | | Flight Systems Report 98-10-24 Revision C dated December 022004 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 29.11.2005 |
| 10015611 | L-3 COMMUNICATIONS / AVIATION | EASA.IM.A.S.01547 | | | DC-9 AND MD-88 | Increasing data recording parameters to 17/18 parameters on digital flight data recorders (DFDR) for L-3 Com (formerly Loral/Fairchild) DFDR and Allied Signal DFDR | | Flight Systems Engineering Inc., Report n° 98-01-05 Rev. J dated 09-5-05 | | 27.01.2006 |
| 10015971 | L-3 COMMUNICATIONS / AVIATION | EASA.IM.A.S.02122 | | | F28 MARK 0100 | Addition of three parameter outputs (Control column, control wheel, and rudder pedals) to the existing Digital flight data recorder(DFDR) to meet requirements for FAR 121.344 (c) and (d) | | Flight Systems Engineering Report for the Additional Parameters for Flight Recording System on Fokker, F-100 Aircraft, L3 Communications Document No. FSE 98-06-01, Revision C. | 1. n/a | 20.12.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|---------------------|----------------|----------------------|---|--|--|--|--|---------------|
| 10016509 | L-3 COMMUNICATIONS / AVIATION | EASA.IM.A.S.02876 | | FAA STC ST03560AT | DC-9-81, DC-9-82, DC-9-83, DC-9-87, MD-88 | Installation of L3 Communications Aviation Recorder Mirco Quick Access Recorder (uQAR). | | Master Drawing List No. 905-E4310-20 Rev. D dated 26.02.2008. Instructions for continued Airworthiness No. 905-E4311-20 Rev. A Dated 23.10.2007 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 21.07.2008 |
| 10016133 | L-3 COMMUNICATIONS INTEGRATED | EASA.IM.A.S.02329 | | FAA STC ST9765SC-D | BOEING 767 | Installation of a First-Class Interior/The project installs an all first-class interior in a Swiss registered Boeing 767 aircraft. This interior consists of installing 54 seats, a bar and dining area, and upgrades to entertainment system electronics. | | The installation of a First-Class Interior in accordance with: L-3 Communications Integrated Systems L.P. Drawing List Section 2.1 of Report 33263-00004, Revision A, dated May 1, 2007. L-3 Airplane Flight Manual Supplement, dated May 1, 2007. | 1. Instruction for Continued Airworthiness (ICA) dated May 1, 2007.2. This approval is limited to only the installation made in the Boeing Model B767-306, Serial Number 30393 aircraft. | 15.05.2007 |
| 10016313 | L-3 COMMUNICATIONS INTEGRATED | EASA.IM.A.S.02617 | | FAA STC ST09101AC-D | BOEING 737-700 IGW | Modification of aircraft environmental control system. | | Raytheon Systems Company 'Type Design Data List' Report no.98010304-000. | Raytheon Systems Company 'Type Design Data List' Report no.98010304-000. | 11.02.2008 |
| 10016314 | L-3 COMMUNICATIONS INTEGRATED | EASA.IM.A.S.02618 | | FAA STC ST09102AC-D | BOEING 737-700 IGW | Installation of passenger gaseous oxygen equipment. | | Raytheon Systems Company 'Type Design Data List' Report no.98010204-000/ | Raytheon Systems Company 'Type Design Data List' Report no.98010204-000. | 11.02.2008 |
| 10016854 | L-3 WESCAM INC. | EASA.IM.R.S.01230 | REV. 1 | TCCA STC SH04-29 | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350N | MX-15 camera installation (structural provisions only). Note: The revision extends the applicability of STC EASA.IM.R.S.01230 to AS355N. | | Installation/Operating Data.# Installation must be done in accordance with DECA Aviation Engineering Modification Summary MS02146, Revision 3, or later approved revisions.# The installation must be maintained in accordance with DECA Aviation E (...) | | 21.01.2008 |
| 10017071 | L-3 WESCAM INC. | EASA.IM.R.S.01507 | REV. 1 | TCCA STC SH04-29 | SEE EASA APPROVED MODEL LIST | MX-Series Camera Installation (Structural Provisions Only) Note: This revision to the STC adds a second Downlink antenna (OMNI Antenna) and a MX-10 Camera applicability. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FMS02146 revision 3 dated 4 October 2010 - Flight Manual Supplement or later revisions of the above listed documents approved by EASAMS02146 revision 6 dated 4 October 2010 - Engineering Modification Summary CA02146 revision 3 dated 4 October 2010 - Instructions for Continued Airworthiness | This STC is for structural provisions only. Functional and electrical aspects will require separate airworthiness approval. VNE limited to 126kts VFR Only Category A operations prohibited | 14.01.2011 |
| 10027823 | L-3 WESCAM INC. | P-EASA.IM.R.S.01517 | REV. 1 | TCCA STC SH97-31 | 1. AS350 B, B1, B2, B3, BA, D 2. AS355 E, F, F1, F2, N, NP | MX-Series Camera Installation (Structural Provisions Only) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | FMS02146 revision 3 dated 4 October 2010 - Flight Manual Supplement or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision) MS02146 revision 6 dated 4 Oc (...) | This STC is for structural provisions only. Functional and electrical aspects will require separate airworthiness approval. VNE limited to 126kts VFR Only Category A operations prohibited | 04.01.2011 |
| 10015759 | LAKE CENTRAL AIR SERVICES | EASA.IM.A.S.01795 | | TCCA STC SA-05-71 | LAKE LA 4/200 | Horizontal Stabilizer Attachment Fitting P/N LC-2200-21 TCCA STC SA-05-71 issued on 8th July 2005 | | - Drawing LC-2200-21, Rev. 0, dated July 06, 2005. - Manufacturing Procedures MP-LC-2200-21, Rev. B, dated July 06, 2005, or later approved revisions. - Services Installation Instructions CI-LC-220-21, Rev. A, dated June 16, 2005, or later approved re (...) | - Parts must be manufactured in accordance with Lake Central Air Services Drawing LC-2200-21, Rev. 0, dated July 06, 2005, and Lake Central Air Services Manufacturing Procedures MP-LC-2200-21, Rev. B, dated July 06, 2005, or later approved revisions. (...) | 18.04.2006 |
| 10043079 | LAKE CENTRAL AIR SERVICES | EASA.IM.A.S.02885 | | TCCA STC SA06-47 | 208 SERIES | Installation of externally mounted survey equipment as per Transport Canada STC SA06-47 | | -, Installation to be accomplished in accordance with Lake Central Air Services documents: -, LCA00304-MP: C-208 series Tail Boom Manufacturing Procedure, Revision 1, dated April 29, 2008, or later Transport Canada approved revision: -, LCA00318-MP: C-208 (...) | none | 18.05.2009 |
| 10044488 | LAKE CENTRAL AIR SERVICES | | REV. 1 | TCCA SA08-85 | SEE EASA APPROVED MODEL LIST | Installation of internally mounted survey and research equipment and supporting systems in accordance with TCCA STC SA08-85. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Instructions law. Eligibility List LCA-SFW-2, Rev. 26, dated 19 July 2010 (Instructions for Continued Airworthiness Doc. No. LCA00610-CD Rev. 0) or later revisions of the above listed documents approved by EASA in accordance with the Technica (...) | 1. Unless otherwise specified, this approval includes the activation of all electrical components of the survey system (re. also SA06-47). 2. Applicable only to Beech B200 Series: The approval in this case covers only the structural provisions that inc (...) | 25.04.2013 |
| 10015401 | LANDMARK AVIATION | EASA.IM.A.S.01111 | | FAA STC ST451CH-D | CESSNA 650 | Installation of an AlliedSignal Avionics MK-VII, Enhanced Ground Proximity Warning System | | -FAA STC ST451CH-D-Garrett Aviation Services drawing list DL541 revision B dated August 12 1998 or later approved revision-1/AFM Supplement dated August 21 1998 or later approved revision-2/AFM Supplement AFMS202 dated January 23 1999 or AFMS204 da (...) | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 24.03.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|-------------------|----------------|----------------------|--|--|--|--|---|---------------|
| 10015402 | LANDMARK AVIATION | EASA.IM.A.S.01112 | | FAA STC ST451CH-D | CESSNA 650 | Installation of a Bendix/King CAS 67A TCAS II System with Dual Bendix/King MST-67A Mode S Transponders | | -FAA STC ST451CH-D-Garrett Aviation Services Drawing List DL38102 no revision dated January 19 1995 or later approved revision- 1/AFM Supplement dated February 14 1995 no revision or later approved revision- 2/AFM Supplement document 53-8783-001 no re (...) | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 24.03.2005 |
| 10015814 | LANDMARK AVIATION | EASA.IM.A.S.01884 | | SA00987CH | CESSNA 525 | Installation of an Allied Signal two tube EFIS-50 on Co-pilot's panel with Honeywell FZ500 Flight Director in accordance with ElectroSonics Drawing No. ES525-720250-01, Revision 1, dated February 2, 1999 or later FAA approved revision. | | see above | 1. Compatibility of this design change with previously approved modifications must be determined by the installer.2. The aircraft must be maintained in accordance with FAA Approved/Maintenance Manual Supplement, document number ES525-720250-200, Init (...) | 05.10.2006 |
| 10016028 | LANDMARK AVIATION | EASA.IM.A.S.02197 | | ST00939CH-D | FALCON 10 | Validation of FAA STC ST00939CH-D: Installation of a Honeywell Enhanced Ground Proximity Warning System (EGPWS) in accordance with Garrett Aviation Drawing List DL939 Revision F, dated 20th January 2005. | | See-FAA STC ST00939CH-D-Airplane Flight Manual Supplement AFMS201 Rev B for configurations 1,2 and 3-Airplane Flight Manual Supplement 1002826 Rev B for configurations 4 and 6-Airplane Flight Manual Supplement 1003542 Rev A for configuration 5-Maintenance (...) | Compatibility with previous approved modifications shall be determined by the installer.JAA Leaflet TGL 12 Certification Considerations for the Terrain Awareness Warning System: TAWS requisites have been applied.Configurations 2 and 3 (without Terra (...) | 06.02.2007 |
| 10016032 | LANDMARK AVIATION | EASA.IM.A.S.02201 | | FAA STC ST01579CH-D | MYSTERE FALCON 20-C5 MYSTERE FALCON 20-D5 MYSTERE FALCON 20-E5 MYSTERE FALCON 20-F5 MYSTERE FALCON 200 | Installation of a Honeywell MK-VIII Enhanced Ground Proximity Warning System (EGPWS) | | - Garrett Aviation Services drawing List 70-0739-001, Revision -, dated April 11, 2002, or later approved revision;- FAA Approved Airplane Flight Manual Supplement document70-0739-003 revision - dated June 7, 2002, or later approved revision;- FAA Approved Airplane Flight Manual Supplement | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 16.10.2006 |
| 10016037 | LANDMARK AVIATION | EASA.IM.A.S.02206 | | FAA STC ST01579CH-D | MYSTERE FALCON 20-5 MYSTERE FALCON 20-C5 MYSTERE FALCON 20-E5 MYSTERE FALCON 20-F5 MYSTERE FALCON 200 | Installation of a Honeywell MK-VIII Enhanced Ground Proximity Warning System (EGPWS) in accordance with Garrett Aviation Services Drawing List 70-0739-001 revision G dated 05th October 2005 or later approved revisions.EASA Validation of:Federal Aviation Administration (FAA) STC ST01579CH-D | | # FAA STC ST01579CH-D.# Airplane Flight Manual Supplement 70-0739-003 Revision B or later approved revisions for aircraft configuration 01.# Airplane Flight Manual Supplement 69-8039-001 Revision C or later approved revisions for aircraft c (...) | 1. JAA Leaflet TGL 12 Certification Considerations for the Terrain Awareness Warning System: TAWS requisites have been applied.2. Prior to installation of this modification the installer must determine that the interrelationship between this m (...) | 03.07.2007 |
| 10016209 | LANDMARK AVIATION | EASA.IM.A.S.02455 | | FAA STC ST01136CH-D | | Installation of a Senior Flexonics Inc. Signal Conditioner Unit | | Garrett Aviation Services drawing list 38-0136-001, dated January 28, 2000, Revision 0, or later approved revision. | Compatibility of this design change with previously approved modifications must be determined by the installer.If the holder agrees to permit another person to use this certificatedo alter the product, the holder shall give the other person evidence of that permission. | 31.05.2007 |
| 10045970 | LANITZ-PRENA FOLIEN FACTORY GmbH | | | | DR 300/108, DR300/180 R DR 300/120 DR 300/140, DR 300/125 DR 340, DR 315, DR 360, DR 380 DR 400 RP, DR 400 NGL DR 400/120, DR 400/125I DR 400/125, DR 400/140 DR 400/140 B, DR 400/120A DR 400/160 D, DR 400/120 D DR 400/160, DR 400/180 DR 400/180 R, DR 400/2+2 DR 400/180 S, DR 400/100 DR 400/200 R, DR 400/500 | Application of ORATEX aircraft covering system | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:CS-23, Amct. 3, for the areas affected by the change;plus for guidance ETSO C14b & ETSO C15d)The requirements for environ (...) | Application of ORATEX covering & coating system, Doc. No.ADXC-51-CO-001, issue C, dated 09-Apr-2013ORATEX Application Manual & Airplane Maintenance Manual Supplement, Doc.No. ADXC-51-001-AMM, Issue 1, dated 13-Jun-2013or later revisions of the above listed documents approved by EASA | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the productT (...) | 01.08.2013 |
| 10028014 | LAST REFUGE Ltd. | P-EASA.A.S.03964 | REV. 1 | | FR172J, R172K, 182, 206, 208 | Approval of Flight Manual Supplements for Last Refuge Ltd Wing Strutmounted camera installation. | BCAR Section K | Flight Manual Supplements:-,LAST REFUGE Ltd. WING STRUT MOUNTED CAMERA SYSTEM (MARK II) FORCESSNA AIRCRAFT , SUPPLEMENT to CESSNA FR172J Reims Rocket, WITHCONTINENTAL IO-360 210HP, FLIGHT MANUAL, AAE SUPPLEMENT 1, ISSUE 2,dated 20 August 2009, LA (...) | None | 25.11.2009 |
| 10028910 | LATITUDE TECHNOLOGIES CORPORATION | | | | 212, 412 | Skynode S200-xxx SATPHONE/TRACKER system product design in accordance with Latitude Technologies Corporation Skynode S200-xxx Master DocumentList No S200-000-001 Revision 1.09, dated Aug 10, 2009. | As defined in FAA TCDS HHSWThe Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | - Latitude Technologies Corporation Skynode S200-xxx Rotorcraft FlightManual Supplement No S200-402-002, Revision 01, (EASA specific RFMSupplement) or later revisions of the abovelisted documents approved by EASA- Latitude Technologies Corporation S (...) | None | 18.02.2010 |
| 10030640 | LATITUDE TECHNOLOGIES CORPORATION | | | TCCA STC SH10-16 | AS355E, AS355F, AS355F1, AS355F2, AS355N, AS355NP | Installation of Latitude Technologies S100 Tracking System | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval | 1117-MDL-001 revision N/C dated 7 April 2010 - Master Drawing List;1117-400-004 revision 01 dated 7 April 2010 - Instructions for ContinuedAirworthiness | None. | 29.06.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|------------------|----------------|----------------------|--|---|--|--|---|---------------|
| 10030644 | LATITUDE TECHNOLOGIES CORPORATION | | REV. 1 | TCCA STC SH10-16 | 206, 206A, 206A-1, 206B 206B-1, 206L, 206L-1 206L-3, 206L-4, 407 | Installation of Latitude Technologies S100 Tracking System | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 1117-MDL-001 revision N/C dated 7 April 2010 - Master Drawing List; 1117-400-004 revision 01 dated 7 April 2010 - Instructions for Continued Airworthiness | None | 14.07.2010 |
| 10030765 | LATITUDE TECHNOLOGIES CORPORATION | | | TCCA STC SH10-16 | AS350B, AS350B1, AS350B2, AS350B3, AS350BA, AS350D, AS350D1 | Installation of Latitude Technologies S100 Tracking System | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 1117-MDL-001 revision N/C dated 7 April 2010 - Master Drawing List; 1117-400-004 revision 01 dated 7 April 2010 - Instructions for Continued Airworthiness | None | 07.07.2010 |
| 10038140 | LAZ-TEC, L.L.C. | | | FAA STC SR00716DE | BELL 429 | Crew Door Assist Handle | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AA-10134 revision A dated 9 February 2011 - Aeronautical Accessories Drawing List AA-10133 revision IR dated 21 December 2010 - Instructions for Continued Airworthiness. | None | 23.01.2012 |
| 10014950 | LD AVIATION PRAGUE S.R.O. | EASA.A.S.03213 | | | CESSNA T303 | COMM/NAV/GPS GNS 530, GNS 430, Navigation Indicator GI 106.S-mode GTX 330 Transponder, GMA 340 ICS, and ELT Kannad 406AF installation into Cessna T303. | | Supplement to Airplane Flight Manual No. 7, 8, 9, 10, 11, 12 ref. S7 LDA 02/07/T-303 revise 3S8 LDA 02/07/T303 revise 3S9 LDA 02/07/T303 revise 3S10 LDA 02/07/T303 revise 3S11 LDA 02/07/T303 revise 3S12 LDA 02/07/T303 revise 3 | S/N: T303 00205. | 15.11.2007 |
| 10014958 | LD AVIATION PRAGUE S.R.O. | EASA.A.S.03228 | | | CESSNA T206H | Installation of Leica RC30 with IPAS 10 and FCMS options. | | Supplement to Airplane Flight Manual No. 9 (ref. S9 LDA/04/07/T206H) | S/N T20608636. | 19.10.2007 |
| 10026761 | LD AVIATION PRAGUE S.R.O. | P-EASA.A.S.03837 | | | CESSNA 172 RG | Avionics Upgrade - GNS430W, KX165A, GMA340, G1106A, KN62A, GTX330, KANNAD406AF Compact Installation | FAR Part 23, Amendment 49 dated 11 March 1996 | Flight manual supplement:..., S21 LDA03/09/C172RGGNS 430 W VHF COMM Transceiver, VOR/ILS and GPS Receiver, S22 LDA 03/09/C172RG KX 165A VHF COMM TRANSCEIVER/ NAV RECEIVER, S23 LDA03/09/C172RG GI 106A NAV indicator, S24 LDA 03/09/C172RG GMA 340 AUI (...) | None. | 07.08.2009 |
| 10036826 | LD AVIATION PRAGUE S.R.O. | | | | L 200A, L 200 D | Avionics Upgrade - GNS 430W, GMA 347, WX 500, KCS 55A, ELT406, MID4200-10 Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. FAR Part 23 Amdt. 49, dated 11 March 1996. The requirements for environmental protection (...) | S1-LDA01/11/L200 COMMNAV/GPS GARMIN GNS 430W, Rev.0-S2-LDA01/11/L200 AUDIO PANEL GARMIN GMA 347, Rev.0-S3-LDA01/11/L200 SYSTEM GYROMAGNETICKÉHO KOMPASU A HSI KC55A, Rev.0-S4-LDA01/11/L200 UMĚLY HORIZONT MID 4200-10, Rev.0-S5-LDA01/11/L200 STORMISC (...) | None | 10.10.2011 |
| 10029813 | LE BOZEC FILTRATION & SYSTEMS SAS | | REV. 2 | FAA STC ST09864SC | 737-100, -200, -200C, -300 737-400, -500 | Rain Repellent System Modification - Modification of Rain Repellent System in accordance with 'Le Bozec, Inc.# documents | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | B737 series Rain repellent Installation-Top Drawing List - Ref. 990110000 Rev.A dated 18/04/2001 - Engineering order - Ref.99010800 Rev.A dated 11/04/2001 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Dec (...) | Only applicable to Rain repellent systems that were installed at time of airplane production under TC EASA I.M.A.120 or FAA TC A16WE. | 06.08.2012 |
| 10031318 | LEARJET INC | | | FAA STC SA8119NM-D | CL-600-2B16 | Installation of the Bulkhead F.S.353 | EASA IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | FAA STC No. SA8119NM-D dated 24 June 1998.Learjet Master Drawing List No. CA-3059a-TUC Revision (-) dated 04 May 1998 or later approved revision or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | none | 10.08.2010 |
| 10031319 | LEARJET INC | | | FAA STC SA8120NM-D | CL-600-2B16 | Interior Installation - Main Cabin | EASA IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | FAA STC No. SA8120NM-D dated 25 June 1998.Learjet Master Drawing List No. CA-3003F-TUC Revision B dated 24 June 1998 or later approved revision or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | none | 10.08.2010 |
| 10031320 | LEARJET INC | | | FAA STC SA8122NM-D | CL-600-2B16 | Installation of the Armledge / Dado | EASA IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | FAA STC No. SA8122NM-D dated 24 June 1998.Learjet Master Drawing List No. CA-3065F-TUC revision A dated 10 June 1998 or later approved revision or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | none | 10.08.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------|-------------------|----------------|----------------------|--|--|--|--|--|---------------|
| 10015312 | LEARJET INC. | EASA.IM.A.S.00162 | | FAA STC ST01606LA-D | BOMBARDIER BD-100-1A10 (CHALLE) | The modification consists in the installation of an Executive Interior in accordance with Master Drawing List HC200135001, Rev. E dated July 10, 2004 (or later FAA approved revision). Validation of FAA STC ST01606LA-D. | | | 1. The Maintenance Instructions for Continued Airworthiness are contained in Documents Number HC200135000-IFCA-1. Modification of an aircraft by this STC obligates the aircraft operator to include the maintenance information provided by this document (...) | 07.10.2004 |
| 10015329 | LEARJET INC. | EASA.IM.A.S.00322 | | FAA STC ST00151BO | LEARJET 55 | Installation of Honeywell CAS-67A TCAS II Software Change 7.0. | | | 1. FAA STC ST00151BO.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 09.11.2004 |
| 10015330 | LEARJET INC. | EASA.IM.A.S.00323 | REV. 5 | FAA STC SA8168NM-D | CL-600-2B16 (CL-604) | Installation of a Complete Custom Aircraft Interior | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Modification Applicability List CC-0001-TUS, Revision DM dated 23 September 2008-MDL CC54495000, Revision A, (for MSN 5449)-MDL DC55105000, Revision B, (for MSN 5510)-MDL DC55235000, Revision E, (for MSN 5523)-MDL DC55395000, Revision E, (for MSN 55 (...)) | Restricted to MSN 5449, 5510, 5523, 5539, 5572, 5577, 5585 | 04.03.2010 |
| 10015357 | LEARJET INC. | EASA.IM.A.S.0065 | | FAA STC ST01595LA-D | BOMBARDIER BD-100-1A10 (CHALLE) | The modification consists in the installation of Control Wheel Lighted Chart Holders, in accordance with Master Drawing List HC33105000, Revision B, dated April 23, 2004 or later FAA approved revision. This STC is the validation of the FAA approved STC No. ST01595LA-D | | | 1. Modification of an aircraft by this supplemental TypeCertificate obligates the aircraft operator to include the maintenance information provided by Document Number HC33105000-IFCA-1 in the operator's Aircraft Maintenance Manual and the operator (...) | 28.07.2004 |
| 10015359 | LEARJET INC. | EASA.IM.A.S.0066 | | FAA STC ST01603LA-D | BOMBARDIER BD-100-1A10 (CHALLE) | The modification consists in the installation of a Service Panel Lighting System, in accordance with Master Drawing List HC33315000, Revision A, dated June 03, 2004 or later FAA approved revision. This STC is the validation of the FAA approved STC No. ST01603LA-D. | | | 1. Modification of an aircraft by this Supplemental TypeCertificate under -1 Configuration/Installation, obligates the aircraft operator to include the maintenance information provided by Document Number HC33315000-IFCA-1 in the operator's Aircraft Maintenance Manual and the operator (...) | 28.07.2004 |
| 10015360 | LEARJET INC. | EASA.IM.A.S.0067 | | FAA STC ST01253LA-D | BD-100-1A10 (CHALLENGER 300) | The modification consists in the installation of Cockpit Writing Tables, in accordance with Master Drawing List HC31515000, Revision D, dated December 18, 2003 or later FAA-approved revision. This STC is the validation of the FAA approved STC No. ST01253LA-D. | | | 1. Modification of an aircraft by this Supplemental TypeCertificate obligates the aircraft operator to include the maintenance information provided by Document Number HC31515000-IFCA in the operator's Aircraft Maintenance Manual and the operator (...) | 28.07.2004 |
| 10015365 | LEARJET INC. | EASA.IM.A.S.0068 | | FAA STC ST01254LA-D | BOMBARDIER BD-100-1A10 (CHALLE) | The modification consists in the installation of an Artex ELT with Nav Interface in accordance with Master Drawing List HC25615000, Rev. D dated December 19, 2003 (or later FAA approved revision). This STC is the validation of the FAA approved STC No. ST01254LA-D. | | | 1. The Maintenance Instructions for Continued Airworthiness are contained in Documents Number HC25615000-IFCA. Modification of an aircraft by this STC obligates the aircraft operator to include the maintenance information provided by this document (...) | 28.07.2004 |
| 10015392 | LEARJET INC. | EASA.IM.A.S.01091 | | FAA STC ST01619LA-D | BOMBARDIER BD-100-1A10 (CHALLE) | Installation of an Executive Interior in accordance with Master Drawing List HC200335000 Revision D or later FAA approved revision | | FAA approved AFM supplements H-FM25285002 rev A dated 29 July 2004. | 1. The Maintenance Instructions for Continued Airworthiness are contained in Document Number HC200335000-IFCA-1. Modification of an aircraft by this STC obligates the aircraft operator to include the maintenance information provided by this document (...) | 24.01.2005 |
| 10015486 | LEARJET INC. | EASA.IM.A.S.01247 | | FAA STC ST01720LA-D | LEARJET 31 LEARJET 35 LEARJET 35A LEARJET 36A | Dual Honeywell MST-67A Diversity Transponders for Elementary & Enhanced Surveillance | | Master Document List BAS35010075 revision F dated April 6, 2005 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 26.07.2005 |
| 10015532 | LEARJET INC. | EASA.IM.A.S.01412 | | FAA STC ST00190DE-D | CL-600-2B16 | Installation of Honeywell FMZ-2000 Flight Management System (FMS) with a Honeywell GNSSU 12 Channel GPS Sensor | | Learnjet Master Document List 7502481 revision (-) dated October 17, 1995 or later approved revision AFM Supplement report number BAS63040001 revision (-) dated October 27, 1995 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 26.07.2005 |
| 10015534 | LEARJET INC. | EASA.IM.A.S.01418 | | FAA STC ST01235LA-D | BOMBARDIER CL-600-2B16 | Universal Avionics TT-3000 AERO-M Satcom System. Master Drawing List No. DC 23155002 Revision E or later FAA approved revision | | | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 09.08.2005 |
| 10015535 | LEARJET INC. | EASA.IM.A.S.01419 | | FAA STC SA8195NM-D | BOMBARDIER CL-600-2B16 | Observer Audio Panel Installation. Master Drawing List No. DC23515000 Revision F or later FAA approved revision. | | FAA Airplane Flight Manual Supplement listed in the approval Master Drawing List is required as part of the installation. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 09.08.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------|-------------------|----------------|----------------------|--|---|---------------------|--|---|---------------|
| 10015536 | LEARJET INC. | EASA.IM.A.S.01420 | | FAA STC ST01258LA-D | LEARJET 45 | Installation of a Universal UMS-1C+ Flight Management System Upgrade from Software Control Number (SCN) 602.X to (SCN) 80X.X with Data Transfer Unit | | Learjet Inc Master Document List BAS45010016 revision C dated March 16 2004 or later approved revisions.AFM Supplement BAS45040103 revision NC dated March 16 2004 or later approved revision is required for the - 1 Configuration.AFM Supplement BAS450 (...) | 1. Universal UNS-1C FMS System must be installed prior to installation of this STC.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with (...) | 18.08.2005 |
| 10015537 | LEARJET INC. | EASA.IM.A.S.01421 | | FAA STC ST00412DE-D | LEARJET 45 | Installation of UNS-1C Flight Management System (Software Version 604.X) | | Learjet Inc Master Document List BAS45010006 revision (-) dated April 7 2000 or later approved revisions.AFM SupplementBAS45040100 revision (-) dated April 07 2000 or later approved revision. | 1. Instructions for Continued Airworthiness BAS-45010006-IFCA or BAS-45010006-IFC-2 dated March 28 2000 or later approved revision required for this installation.2. This STC is approved only for the product configuration as defined in the appro (...) | 18.08.2005 |
| 10015549 | LEARJET Inc. | EASA.IM.A.S.01443 | | FAA STC ST01463SE | LEARJET 31 | Group Aircraft Approval for Reduced Vertical Separation Minimum (RVSM) | | - Learjet Master Drawing List (MLD) BAS31015030 Revision N/C dated April 15, 2005 or later approved revision.- AeromechAFM Supplement AMI-AFMS-0428 Rev A approved May 09 or later approved revision.- Instructions for initial & Continued Airworthiness AMI-IICA-LJ-0432 Revision B dated May 5, 2005 or | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 08.12.2005 |
| 10015616 | LEARJET INC. | EASA.IM.A.S.01555 | | FAA STC SA8171NM-D | BOMBARDIER BD-700-1A10 | Pilot and Co-Pilot Footrests for BF700-1A10 Global Expressacc.: MDL GC2500501 Rev. B or later FAA approved rev.Validation of imported FAA STC: SA8171NM-D | | | | 16.11.2005 |
| 10015710 | LEARJET Inc. | EASA.IM.A.S.01703 | | FAA STC ST01181WI-D | LEARJET 31 LEARJET 35 LEARJET 35A LEARJET 36 LEARJET 36A | Installation of Honeywell IHAS-800 Multi-Hazard Awareness System | | *Learjet Master document List BAS35015037 Revision L or later approved revision-configuration | Learjet Master Document List BAS35015037 Revision I or later approved revision - configurations -2, -4 and -5 only | 22.03.2006 |
| 10015730 | LEARJET INC. | EASA.IM.A.S.01749 | | FAA STC ST00342DE-D | LEARJET 31A | Installation of Single Universal Avionics Corporation UNS-1C FMS | | Learjet Inc Master Document List BAS31010008 Rev (-) dated March 5 1998 or later approved revisionAFM Supplement BAS31040105 Revision (-) dated March 5 1998 or later approved revision or Supplement BAS31040008 Revision (-) dated February 16 2000 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 02.01.2006 |
| 10015788 | LEARJET INC. | EASA.IM.A.S.01844 | | FAA STC ST00508DE-D | BOMBARDIER CL-600-2B16 | Installation of Collins PCD-3000 Data Loader Interface System | | Learjet Inc Master Document List BAS63010085 Revision (-) dated January 22, 2002 or later approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 20.01.2006 |
| 10015790 | LEARJET INC. | EASA.IM.A.S.01848 | | FAA STC ST8139NM-D | BOMBARDIER CL-600-2B16 | Installation of a Collins VHF-22C Comm System with CTL-22C. | | # FAA STC No. SA8139NM-D# Learjet Inc. MDL No. BAS63010060, Revision J, dated 9 March 2000 or later approved revision.# Learjet Inc. AFM Supplement No. BAS63040111, Revision A, dated 3 July 1999 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and #Associated Technical Documentation#. Compatibility with other aircraft/engine configurations shall be determined by theinstaller. | 07.05.2007 |
| 10015796 | LEARJET INC. | EASA.IM.A.S.01855 | | FAA STC ST01717LA-D | BOMBARDIER BD-100-1A10 (CHALLE) | installation of a lavatory Seat Modification - Belt replacement, in accordance with Master Drawing List BAS68010002, revC | | Maintenance Instructions for Continued Airworthiness ref BAS68010002-IFCA-1 (Learjet CRI D-3) | Certification basis: a deviation from 25.789(b) has been granted for personal injury criteria testing of side single occupancy seats | 07.04.2006 |
| 10015817 | LEARJET INC. | EASA.IM.A.S.01888 | REV. 1 | SA06-10 | BOMBARDIER BD-700-1A10 BOMBARDIER BD-700-1A11 | EGPWS Software Upgrade Installation of Cockpit Speaker Mute | | FAA SOC dated 06 May 2008 and Bombardier Aerospace BusinessAviation Services Master Document List, document number BAS87017015, Revision F, dated May 04, 2007 or later approved revision. | 1. The approval for the Master Document List -1 configuration, EGPWS Software Upgrade (218-218 Software), is contingent on the aircraft being equipped with a previously EASA approved installation of a MK V Honeywell EGPWS with Version -206-206 Soft (...) | 03.06.2008 |
| 10015852 | LEARJET Inc. | EASA.IM.A.S.01955 | | FAA STC ST01186WI-D | LEARJET 31 LEARJET 35 LEARJET 35A LEARJET 36 LEARJET 36A | Installation of Universal UNS-1L Flight Management System. | | FAA STC ST01186WI-DLearjet Master Document List BAS35010062Revision H dated September 10 2004 or later approved revision.AFM Supplement ref. BAS35040122 Revision (-) dated February 19 2004 or later approved revision. | Not JAA TGL10 Compliant with present software load. | 13.10.2006 |
| 10015884 | LEARJET INC. | EASA.IM.A.S.02002 | | FAA STC SA8197NM-D | BOMBARDIER CL-600-2B16 (604) | Installation of Oxygen System-Passenger & crew (14500ft) | | Learjet Inc Master Document List DC35205000 Revision NC dated April 18, 2002 or later approved revision.AFM Supplement ref D-FM35205002 Revision NC dated March 12, 2002 or later approved revision. | -- | 20.06.2006 |
| 10015919 | LEARJET INC. | EASA.IM.A.S.02052 | | ST01237LA-D | BOMBARDIER CL-600-2B16 (CL-604) | Installation of a Complete Custom Interior | | FAA STC ST01237LA, DMaster document List DC55325000 RevisionD dated May 23 2003 or alter approved revisionAFM Supplement are listed on Lelectrical Drawing Index TM4500-5117 as listed on approved Master Document List | Modification of an aircraft by this STC obligates the aircraft operator to include the maintenance information TE-55325000-IFCA in the operators Maintenance Manual & the OperatorsScheduled Maintenance ProgramThe equipment for which provisions were p (...) | 03.11.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------|-------------------|----------------|----------------------|---|---|---|--|--|---------------|
| 10015920 | LEARJET INC. | EASA.IM.A.S.02053 | | FAA STC SA8109NM-D | BOMBARDIER CL-600-2B16 | Installation of Honeywell Enhanced Ground Proximity Warning System with Runway Awareness & Advisory System (RAAS) | | FAA STC SA8109NM-D Learjet Inc. Master Drawing List CA-1072-TUC Revision AC or later approved revision. Learjet Inc. AFM Supplement BAS63047012 Revision NC or later approved revision. | | 05.07.2007 |
| 10015948 | LEARJET INC. | EASA.IM.A.S.02090 | REV. 1 | FAA STC SA8129NM-D | BOMBARDIER CL-600-2B16 (604) | Installation of Ultra Electronics Active Noise & Vibration Control system | | FAA STC SA8129NM-D Master Drawing List CA-1077-TUC Revision C dated October 08, 1998 or later approved revision. Learjet Inc AFM Supplement C-1085 Revision A dated November 12 1999 or later approved revision. | | 14.09.2007 |
| 10016066 | LEARJET INC. | EASA.IM.A.S.02245 | | FAA STC ST01731LA-D | LEARJET 45 | Installation of Honeywell Integrated Communication Unit / Enhanced Surveillance Upgrade (-1 Configuration). | | Project Specific Certification Plan BAS45337001, rev. NC. JAR Compliance Checklist BAS45160014, rev. NC. Master Document List BAS45015013, rev. EAFM Supplement BAS45045008, rev. NC. Instructions for Continued Airworthiness BAS45015013-IFCA-1, rev. NC | 1. Incorporation of ECR 7598 and Honeywell Service Bulletins 7517400-23-6008, 7510700-23-6044 and 7510700-23-6046 is a prior or concurrent requirement to this installation. 2. This STC is approved only for the product configuration as defined in the a (...) | 14.03.2007 |
| 10016070 | LEARJET INC. | EASA.IM.A.S.02251 | | FAA STC SA8178NM-D | BOMBARDIER BD-700-1A10 | Installation of an Artex ELT/NAV Interface System. Validation of the FAA STC SA8178NM-D | | IFCA: GC-25615001-IFCA-1 or -2 | | 21.02.2007 |
| 10016106 | LEARJET INC. | EASA.IM.A.S.02296 | | FAA ST00222BO | BOMBARDIER CL-600-2B16 | Installation of enhanced surveillance modification to Dual Collins TDR-94/94D transponder system. | | FAA STC ST00222BO dated 22 November 2005 or later approved revision. BBAS Master Document List BAS 63017005 Revision NC dated 25 october 2005 or later approved revision. BBAS AFM Supplement BAS63047009 Revision NC dated 22 November 2005 or later approved revision. | Installation of enhanced surveillance modification per this STC requires that a previously EASA approved dual installation of Collins p/n 622-9352-007 (TDR-94) or 622-9210-007 (TDR-94D) or later version has been installed in the aircraft. | 31.05.2007 |
| 10016108 | LEARJET INC. | EASA.IM.A.S.02298 | REV. 1 | FAA STC ST01732LA-D | BOMBARDIER BD-700-1A10 | Installation of a Rockwell Collins Integrated Flight Information System (IFIS-5000). | | Master Drawing List, ref. Doc. BAS68010011, Rev. G, dated 28/11/2006. Airplane Flight Manual Supplement, ref. Doc. BAS68040003, Rev. N/C, dated 28/11/2006. Maintenance Requirements - Instructions for Continued Airworthiness, ref. Doc. BAS68010011-IFCA (...) | | 04.12.2007 |
| 10016140 | LEARJET INC. | EASA.IM.A.S.02336 | REV. 3 | FAA STC SA8164NM-D | BD-700-1A10 | Installation of a Complete Custom Aircraft Interior - ASN 9016, 9020, 9054, 9098, 9106 | EASA.IM.A.009 The Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged (...) | -FAA STC No. SA8164NM-D dated 26 August 2002 - Bombardier Master Drawing List No. GC90205000 Revision L dated February 2002 - Airplane Flight Manual Supplements required for this installation are listed on the approved Bombardier Modification Applicability (...) | Restricted to ASN 9016, 9020, 9054, 9098, 9106. The following STC's are a concurrent requirement with this STC: -EASA STC10016535 (FAA STC ST00941LA) - Floor Proximity Emergency Escape Path Marking System is applicable to all aircraft serial numbers apr (...) | 10.08.2011 |
| 10016148 | LEARJET INC. | EASA.IM.A.S.02349 | | FAA STC SA8133NM-D | BOMBARDIER CL-600-2B16 (CL-604) | Replacement of a Collins TCAS II bottom OMNI-Directional Antenna (Sensor Systems) | | FAA STC No. SA8133NM-D Learjet Inc. MDL No. CA-1024C-TUC at Revision A dated 02 February 1999 or later Approved Revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 07.05.2007 |
| 10016160 | LEARJET INC. | EASA.IM.A.S.02382 | | FAA STC ST01082WI-D | LEARJET 60 | Universal UNS-1E/C+ Flight Management System. FAA STC ST01082WI-D | | Design Compliance Report 60-D1810 "MDL M100378ICT" or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | 1. This approval only applies to the -004, -005, -006, and -010 configurations of the Universal UNS-1C+ Flight Management System. 2. See also FAA STC ST01082WI-D3. Prior to installation of this modification the installer must determine that th (...) | 08.08.2007 |
| 10016163 | LEARJET INC. | EASA.IM.A.S.02390 | | FAA STC SA8100NM-D | BOMBARDIER CL-600-2B16 | Installation of an Axle Jack Storage Box Installation | | DAS-12-NM Certification Plan CA-3074-TUC Master Drawing List P030-C-97 Compliance Checklist / MOCSTR-M319 Structures Report | | 31.05.2007 |
| 10016164 | LEARJET INC. | EASA.IM.A.S.02391 | | FAA STC SA8097NM-D | BOMBARDIER CL-600-2B16 | Installation of an Engine Cover Storage - Aft Equipment | | DAS-12-NM Certification Plan CA-3073-TUC Master Drawing List P029-C97 Compliance Checklist / MOCSTR-M311 Structures Report | | 31.05.2007 |
| 10016165 | LEARJET INC. | EASA.IM.A.S.02392 | | FAA STC SA6003NM-D | BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 | Interior Installation/Modification in accordance with DAS approved Learjet Master document list No. CA-1000TUC, Rev "N" dated January 12, 1995 or later approved revision | | Learjet Airplane Manual Supplement C1002, original release, dated March 5, 1993 C1003 original release, dated Febr 23, 1993, C1004, original release, dated Febr 23, 1993, C1005, original release, dated March 10, 1993; C1011, original release, dated Oct 12, 1993 or later approved revision | The proper interrelationship between this installation and any previously approved configuration has to be assured. | 03.08.2007 |
| 10016226 | LEARJET INC. | EASA.IM.A.S.02478 | | FAA STC SA8101NM-D | BOMBARDIER CL-600-2B16 (CL-604) | FAA STC SA8101NM-D, NAV Lights Relocation. This STC supersedes LBA TA0136. | | Bombardier Aircraft Services MDL No. CA-3017-TUC at Revision F dated 05 June 2002 or later approved revision | | 31.05.2007 |
| 10016356 | LEARJET INC. | EASA.IM.A.S.02678 | | FAA SA8191NM-D | BOMBARDIER CL-600-2B16 (604) | Installation of a Safe Flight Enhanced Autopower System. | | Learjet Inc. MDL No. DC22305001 at revision U, dated 26 April 2007. Learjet Inc. AFMS No. BAS63040125 at revision NC, dated 26 April 2007. Required for -3 configuration, or later revisions of the above listed documents approved by EASA in accordance (...) | The -3 configuration requires the installation of FAA STC SA8191NM-D -2 configuration. | 21.08.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------|-------------------|----------------|----------------------|--|---|---|---|--|---------------|
| 10016387 | LEARJET INC. | EASA.IM.A.S.02715 | | FAA STC ST01737LA-D | BOMBARDIER BD-700-1A10 BOMBARDIER BD-700-1A11 | Installation of Honeywell / EMS HD-710 High-Speed Data Satcom System. | | FAA STC No. ST01737LA-D, dated 6 June 2007.FAA Statement of Compliance, dated 7 November 2007.Bombardier Aerospace Master Document List BAS67010038, Revision C, dated 6 June 2007, or later approved revisions.Learjet Inc. Airplane Flight Manual Supplement BAS67040130, Revision NC, dated 6 June 2007. | The installation of an EASA approved Router/LAN System is a prerequisite/ concurrent requirement to the installation of the -1 Configuration. | 07.04.2008 |
| 10016401 | LEARJET INC. | EASA.IM.A.S.02730 | | FAA ST00163BO | BOMBARDIER BD-700-1A10 BOMBARDIER BD-700-1A11 | Installation of a main entry door rain deflector channel. | | FAA STC ST00163BO dated 21 November 2007.FAA SOC dated 06 may 2008.Bombardier Aerospace MDL No. BAS67017003, Revision A, dated 11 November 2007 or later approved revision. | | 08.05.2008 |
| 10016445 | LEARJET INC. | EASA.IM.A.S.02791 | | FAA STC ST02023 LA-D | BOMBARDIER CL-600-2B16 | Installation of a Mid-Cabin Interior Modification. | | Bombardier Aerospace Master Document List BAS63010121, Revision "F", dated February 7, 2008 or later approved revision.Installation for Continued Airworthiness (ICA) for the -1 Configuration/Installation are contained in Document Nnumber BAS63010121- (...) | A prior or concurrent requirement for the installation of the -1 Configuration is the incorporation of EASA STC No. EASA.IM.A.S.02392 for the interior/Modification Installation. | 07.05.2008 |
| 10016481 | LEARJET INC. | EASA.IM.A.S.02834 | | FAA STC ST01730LA-D | BOMBARDIER CL-600-2B16 (604) | FAA STC ST01730LA-D, Installation of an Aircell Access II Inruid Satcom System. | | FAA STC ST01730LA-D dated 15 August 2006.Bombardier Aerospace Master Document List BAS63017007, Revision B, dated August 15, 2006, or later Approved revision.The Maintenance Instructions for Continued Airworthiness (IFCA) for the -1 Configuration / (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 24.06.2008 |
| 10016570 | LEARJET Inc. | EASA.IM.A.S.02967 | | FAA STC ST00994WI-D | LEARJET 60 | Cockpit Speaker Mute FunctionFAA STC ST00994WI-D | | As per Master Document List M100339ICT, rev. G (or later approved revisions) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 06.10.2008 |
| 10016574 | LEARJET Inc. | EASA.IM.A.S.02972 | | FAA STC ST01421WI-D | LEARJET 60 | Airshow 410 System | | As per Master Data List M100475ICT, rev. P (or later approved revisions). | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.The i (...) | 16.09.2008 |
| 10016575 | LEARJET Inc. | EASA.IM.A.S.02973 | | FAA STC ST01424WI-D | LEARJET 60 | Emergency power supply 1 | | As per master Data List M100487ICT, rev. M (or later approved revisions). | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.The i (...) | 16.09.2008 |
| 10016600 | LEARJET INC. | EASA.IM.A.S.03013 | | FAA STC ST10380SC | BOMBARDIER BD-700-1A10 | Replacement of Low Back Seats with High Back Seats. | | - FAA SoC dated 22 December 2008.- FAA STC No. ST10380SC Issue 1 dated 14 February 2005 - Learjet Inc. Master Document List, Document Number BAS67010032, Revision NC, dated 11 February 2005 or later approved revision.- Learjet Inc. Airpl (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 23.12.2008 |
| 10016611 | LEARJET Inc. | EASA.IM.A.S.03031 | | FAA STC ST01370WI-D | LEARJET 60 | Honeywell HF Communication System. | | As per Master Document List M100457ICT, rev. V or later revisions approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision). | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 27.10.2008 |
| 10016618 | LEARJET Inc. | EASA.IM.A.S.03049 | | FAA STC ST01402WI-D | LEARJET 60 | Collins Pro Line 21 Avionics InstallationFAA STC ST01402WI-D | | As per Master Data List AAC-0497-001, rev. L (or later approved revisions) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.The i (...) | 13.11.2008 |
| 10016619 | LEARJET Inc. | EASA.IM.A.S.03050 | | FAA STC ST01422WI-D | LEARJET 60 | Interior Upgrade (Floor Plans A, B, C & D)FAA STC ST01422WI-D | | As per Master Document List M100471ICT, rev. BH (or later approved revisions) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.The installation of this modification by third perso | 13.11.2008 |
| 10016651 | LEARJET Inc. | EASA.IM.A.S.03099 | REV. 2 | FAA STC SA8199NM-D | BD-700-1A10 | FAA STC SA8199NM-D, Installation of a Full Authority Cabin Temp Control. | EASA TCDS EASA.IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product (...) | -FAA STC No. SA8199NM-D dated 01 May 2002.-FAA SoC reference 150L-08-394 dated 10 December 2008.-Bombardier Master Document List No. BAS67010007, Revision D, dated 24 August 2004 or later approved revision.-Bombardier AFMS No. BAS67040007, Revision N (...) | None | 16.12.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------|---------------------|----------------|----------------------|--|--|--|---|--|---------------|
| 10016682 | LEARJET Inc. | EASA.IM.A.S.03205 | | FAA STC ST00604W1 | LEARJET 31A | Installation of stand-alone Garmin GPS-165 Global Positioning System (VFR Navigation Only). | | # FAA STC ST00604W1# Master Document List, ref. doc. No. M100229ICT, Rev. B, dated 2 December 1997 or later revisions approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision)# Airplane Fig (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.The (...) | 28.04.2009 |
| 10027108 | LEARJET INC. | P-EASA.IM.A.S.03188 | | FAA STC ST01716LA-D | BD-100-1A10 (CHALLENGER 300) | - Installation of a Passenger Seat Upgrade-Validation of FAA STC ST01716LA-D | Certification Basis as per EASA TCDS (IM.A.080 Issue 2, 8th October 2008)(JAR 25 Change 15 plus other airworthiness requirements, SpecialConditions and Equivalent Safety Findings) | - FAA STC ST01716LA-D- Master Drawing List, ref. Doc. BAS68010001, Revision C, dated 28thFebruary 2005- Maintenance Requirements - Instruction for Continued Airworthiness.ref. doc. BAS68010001-IFCA-1, dated 24th February 2005(*) or later revisions a (...) | None | 07.09.2009 |
| 10028342 | LEARJET INC. | P-EASA.IM.A.S.03206 | REV. 1 | FAA STC ST01255LA-D | BD-100-1A10 (CHALLENGER 300) | Installation of a Baseline Executive Interior.Validation of FAA STC ST01255LA-D 2 Configuration)Revision of the EASA STC 10028342 to include -2 Configuration | Certification Basis as per EASA TCDS (IM.A.080 Issue 2, 8th October 2008)plus:The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalCRI D-07 Is (...) | - FAA STC ST01255LA-D issued 20th December 2003- Master Drawing List ref. doc. HC200065000 Revision U (-2Configuration), dated 18th September 2004- Airplane Flight Manual Supplement ref. doc. H-FM25285002, Revision A,dated 29th July 2004orlater rev (...) | Not Applicable | 15.03.2010 |
| 10031271 | LEARJET Inc. | | | FAA STC SA8117NM-D | CL-600-2B16 | Installation of the RH Forward Lavatory | EASA.IM.A.023The Certification Basis for the original product remains applicable tothis certificate/ approvalThe requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areuncha (...) | FAA STC No. SA8117NM-D dated 24 June 1998.Learjet Master Drawing List No. CA-3068e-TUC Revision A dated 10 June1998 or later approved revision.or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions ofthis decision) | none | 06.08.2010 |
| 10031299 | LEARJET Inc. | | | FAA STC SA8118NM-D | CL-600-2B16 | Installation of the RH Forward Galley | EASA.IM.A.023The Certification Basis for the original product remains applicable tothis certificate/ approvalThe requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areuncha (...) | FAA STC No. SA8118NM-D dated 25 June 1998.Learjet Master Drawing List No. CA-3067e-TUC Revision B dated 24 June1998 or later approved revision.or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions ofthis decision) | none | 09.08.2010 |
| 10035643 | LEARJET INC. | P-EASA.IM.A.S.02343 | | FAA STC ST01239LA-D | CL-600-2B16 | FAA STC ST01239LA-D, Complete Custom Aircraft Interior | EASA.IM.A.023The Certification Basis for the original product remains applicable tothis certificate/ approvalThe requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areuncha (...) | FAA STC No. ST01239LA-D dated 02 December 2002.Learjet Inc.Master Document List DC55415000 Revision E dated 02December2002 or later approved revision.Airplane Flight Manual Supplements required for this installation arelisted on the Electrical Draw (...) | Restricted to ASN 5541.A concurrent requirement to the installation of Brunoe Aerospace EASA STC No. 10035636(FAA STC SA6011NM) (ref. EASA PNO010011393) Floor ProximityEmergency Escape Path Marking System. | 06.07.2011 |
| 10036055 | LEARJET INC. | P.EASA.IM.A.S.03253 | | FAA STC ST01735LA-D | BOMBARDIER CL-600-1A11 BOMBARDIER CL-600-2A12 BOMBARDIER CL-600-2B16 | FAA STC ST01735LA-D, Installation of a Kidde Aerospace Smoke Detector inthe Aft Baggage Compartment, as a replacement for Walter Kidde PIN473052. | EASA.IM.A.023 | FAA STC No. ST01735LA-D dated 13 March 2007.Bombardier Aerospace Master Document List BAS6301 0115 Revision A dated13 March 2007 or later approved revision.Bombardier Aerospace Airplane Flight Manual Supplement BAS63040218Revision NC dated13 March 2 (...) | An EASA approved Walter Kidde Smoke Detector System for PIN473052 is prerequisite I concurrent requirement for the installation of this STC. | 02.07.2009 |
| 10037572 | LEARJET INC. | | REV. 1 | FAA STC ST00288BO | BD100-1A10 (CHALLENGER 300) | Installation of Honeywell EGPWS with windshear and Terrain Display (MDL-1 configuration), Runway Awareness and AdvisorySystem (RAAS) (MDL -2configuration) or Runway Awareness andAdvisory System (RAAS) inhibit(MDL-3 configuration) in accordance with (...) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate- CRI F-1 issue 2 dated 29/11/2011The requirements for environmental protection and the associated (...) | - FAA Supplemental Type Certificate ST00288BO dated 3, September 2009- FAA approval of MDL BAS68017001 Revision A datedFebruary 28, 2011- Master Drawing List ref. BA568017001 Revision A dated February 28, 2011- Certification plan for Bombardier Aeros (...) | - Instructions for Continued Airworthiness ref. BAS68017001-ICA-1 rev Adated January 14, 2011 must be made available tothe operator at the time of installation for MDL-1 configuration, EGPWS with Windshear andTerrain Display- Instructionfor Continu (...) | 19.03.2012 |
| 10038287 | LEARJET INC. | | REV. 1 | FAA STC ST00299BO | BD-700-1A10, BD-700-1A11 | Installation of Honeywell Cockpit Displays Upgrade | The Certification Basis (CB) for the original product remains applicableto this certificate/ approval.The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/approval. | -FAA STC No. ST00299BO dated 23 September 2011.-Learjet, Inc. Master Document List, Document No. BAS67010045, RevisionC, dated 22 May 2012.-Learjet, Inc. Airplane Flight Manual Supplement (AFMS) Document No.BAS67040028, Revision B, approved 22 May 2 (...) | The Honeywell DU-875 Multifunction Display (MFD) Expansion Applicationsinclude 'MFD Charts' (date sensitive charts withtownship position) and'MFD Uplink Graphical Weather Map' (XMW)weather. Any other intendedfunction of this equipment willrequire a re-examination of thecertification basis (i.e. MF | 19.03.2013 |
| 10040405 | LEARJET INC. | | | FAA STC SA8192NM-D | BD-700-1A10 | Installation of a Sextant Integrated Electronic Standby InstrumentSystem | EASA.IM.A.009The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunch (...) | FAA STC No. SA8192NM-D dated 02 August 2001.Learjet Master Document List GC34235001 Revision G dated 01 August 2001.Learjet Airplane Flight Manual Supplement G-FM34235001 revision(-)dated 27 July 2001.or later revisions of the above listed documents (...) | none | 02.07.2012 |
| 10040951 | LEARJET Inc. | EASA STC NO. 1426 | REV. 1 | FAA STC ST01122SE-D | LEARJET 35/35A/36/36A | Group Aircraft Approval for RVSM Validation of FAA STC ST01122SE-D dated 14 April 2010.Note: This approval constitutes an update to EASA STC ref. N° 1426 dated2 December 2003 (LBASTC TA0751, dated 27 November 2003) in order toreflect the compati (...) | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | - AeroMech Document AMI-QD-LJ-0213, Rev 'new', dated May 2,2003;- Master Document List BAS35015040, Rev B, dated May 19, 2004;- Airplane Flight Manual Supplement AeroMech document AMI-03-ALJ3536SRev 'original', FAA approved May 8, 2003;- Instruction (...) | The following items listed in the Limitations and Conditions section of the FAA STC ST01122SE-D, dated April 14 2010, are excluded from EASAApproval of this STC, because they are related to a configuration(C-21A) that has not been validatedby EASA'S (...) | 07.08.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------|--------------|----------------|----------------------|----------------------------|--|---|--|--|---------------|
| 10044402 | LEARJET INC. | | | FAA STC ST01605LA-D | BD-100-1A-10 | Installation of an executive interior | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:- Equivalent Safety Finding(s): D-06 issue 3, dated 3/04/2013-D-07 issue 3, dated 3/04/2013-D-09 issue 3, dated (...) | MDL HC200155000 Rev N, dated 25/06/2004-AFM supplements H-FM25285002 Rev NC dated 18/12/2003 and H-FM25285003 Rev NC dated 25/06/2004-ICA200155000-IFCA-1 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Deci (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 10.04.2013 |
| 10044763 | LEARJET INC. | | | FAA STC SA8159NM-D | CL-600-2B16 (CL-604) | Installation of a L/H Fwd Storage Cabinet | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. SA8159NM-D dated 29 April 1999. Bombardier Completion Center Master Drawing List No. CA-3079a-TUC, Revision B dated 26 April 1999, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/C (...) | None | 06.05.2013 |
| 10044765 | LEARJET INC. | | | FAA STC SA8094NM-D | CL-600-2B16 | Installation of Cabinets and Galley | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. SA8094NM-D dated 03 May 1997. Bombardier Completion Center Master Drawing Lists: Revision, Date, CA-3058a-TUC, C, 06 February 1999, CA-3060d-TUC, C, 06 February 1999, CA-3061d-TUC, C, 06 February 1999, CA-3065d-TUC, C (...) | This approval is for installation of structural provisions only. The installation of electrical, plumbing and ECS components must be separately approved. | 06.05.2013 |
| 10044766 | LEARJET INC. | | | FAA STC SA8073NM-D | CL-600-2B16 | Installation of Cabinets and Galley | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. SA8073NM-D dated 14 April 1997. Bombardier Completion Center Master Drawing Lists: Revision, Date, CA-3060b-TUC, D, 16 March 1999, CA-3061b-TUC, G, 16 March 1999, CA-3062b-TUC, F, 16 March 1999, CA-3064b-TUC, F, 16 (...) | This approval is for installation of structural provisions only. The installation of electrical, plumbing and ECS components must be separately approved. | 06.05.2013 |
| 10044834 | LEARJET INC. | | | FAA STC ST00097W1 | CL-600-2A12, CL-600-2B16 | Storage Cabinet, LH Aft | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. ST00097W1 dated 7th October 1996. Learjet Master Document List No. CA-1026-TUC Revision C dated 7th July 1993, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent rev (...)) | This approval is for installation of structural provisions only. The installation of electrical, plumbing and ECS components must be separately approved. Prior to installation of this design change it must be determined that the interrelationship between (...) | 08.05.2013 |
| 10044835 | LEARJET INC. | | | FAA STC ST00046W1 | CL600-2A12, CL-600-2B16 | Installation of Left Hand Forward Wardrobe Cabinet | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. ST00046W1 dated 7th October 1996. Learjet Master Document List No. CA-1022-TUC Revision G dated 27th July 1993, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent re (...)) | This approval is for installation of structural provisions only. The installation of electrical, plumbing and ECS components must be separately approved. Prior to installation of this design change it must be determined that the interrelationship between (...) | 08.05.2013 |
| 10044836 | LEARJET INC. | | | FAA STC ST00044W1 | CL600-2A12, CL-600-2B16 | Pullout Table and Sideledge | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. ST00044W1 dated 7th October 2006. Learjet Master Document List No. CA-1023-TUC Revision E dated 7th July 1993, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent rev (...)) | This approval is for installation of structural provisions only. The installation of electrical, plumbing and ECS components must be separately approved. Prior to installation of this design change it must be determined that the interrelationship between (...) | 08.05.2013 |
| 10044889 | LEARJET INC. | | | FAA STC ST2028LA | BD-700-1A-10, BD-700-1A-11 | Installation Aviation Communication & Surveillance Systems (ACSS) TCAS2000 Computer Unit. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. ST2028LA dated 17 December 2012. Learjet Inc. Master Data List (MDL) BAS67010058, Revision NC, dated 17 December 2012, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subs (...)) | This modification applies to airplanes equipped with the Honeywell/ACCTCAS 2000 system only. The incorporation of Aviation Communication & Surveillance Systems (ACSS) Service Bulletin 8008229-001 (FAA AD2012-02-08 refers), in its entirety with no de (...) | 15.05.2013 |
| 10031894 | LEARJET, INC | | REV. 1 | FAA STC ST02025LA-D | 31A | Installation of Northrop Grumman Dual LITEF LCR-100 AHRSA/ASTC ST02025LA-D (issued March 10, 2010) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 25.1316 Amendment 80FAR 25.1317 Amendment 122. The requirements for environmental pro (...) | Refer to FAA STC ST02025LA-D or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | Previous or concurrent requirement for this installation is an approval of installation of the LCR-93 AHRs in accordance with Learjet ECR 4467. | 20.09.2010 |
| 10035128 | LEARJET, INC | | | FAA STC SA8182NM-D | BD-700-1A10 | Modification of the Forward Water Fill Panel | EASA TCDS IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are (...) | FAA STC No. SA8182NM-D dated 23 January 2001. Bombardier Aerospace Master Drawing List BAS67010003 revision A dated 21 December 2000 or later approved revision or later revisions of the above listed documents approved by EASA in accordance with EASA ED (...) | This installation modifies the existing water fill panel installation installed by Bombardier Inc. Engineering Drawing MDC9000-0142-1 only (Previously approved under EASA STC No. EASA.IM.A.S.02336) | 27.05.2011 |
| 10029566 | LEES AVIONICS Ltd. | | | | COMMANDER 112, 114 | Installation of Garmin G500/G600 PFD/ND | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23EASA GM/G/001 has been applied. The certified noise and/or emissions levels of (...) | Master Data List MDL1097-01 Issue 3- Change Statement CS1097-01 Issue 2- Instructions for Continued Airworthiness ICA1097-01 Issue 1- Flight Manual Supplement AFMS1097-01 Issue 3 or later revisions of the above listed documents approved by EASA | This change is limited to the configuration defined in Change Statement CS1097-01 Issue 2 or later approved revision. | 01.04.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------|----------------|----------------------|---|--|--|---|---|---------------|
| 1003384 | LEES AVIONICS Ltd. | | | | LEARJET 35, LEARJET 35A | Installation of single or dual Universal Avionics UNS-1Lor UNS-1LW FMS (Flight Management System) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: Refer to Certification Plan CP-1159, issue 1, and Compliance Record CR-1159, issue The re (...) | Refer to Master Documents List MDL-1159, issue 1 or later revisions of the above listed documents approved by EASA | Refer to Flight Manual Change Sheet FCMS-1159, issue 1 | 17.12.2010 |
| 10036361 | LEES AVIONICS Ltd. | P-EASA.A.S.02912 | | | 1900, 1900C, 1900D 200, 200C, 200CT, 200T, A200 A200C, A200CT, B200, B200C B200CGT, B200CT, B200GT, B200T | TAWS-B Installation. | CS-23The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to thiscertificate/ approval. | Master Data ListMDL77801 Issue 2 dated 08 December '10.Change StatementCS-778 Issue 2 dated 08 December '10Instructionsfor Continued Airworthiness ICA-778 Issue 1 dated 30 April'10.Flight Manual SupplementAFMS-778-01 Issue 1 dated 30 April '10 or la (...) | None | 02.09.2011 |
| 10036482 | LEES AVIONICS Ltd. | P-EASA.R.S.01567 | | | BELL 412 | Installation of the Spectrolab search light SX16. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.The Applicant applied as voluntary compliance the CS 29-2; for theapplicableparagraphs (...) | Rotorcraft Flight Manual supplement RFMS-1041 iss. 1 for later revisions of the above listed documents approved by EASA.-Lees Certification Plan CP-1041 iss. 1- MDL 1041-01 iss. 1 | None. | 16.09.2011 |
| 10036506 | LEES AVIONICS Ltd. | P-EASA.R.S.01570 | | | BELL 412 | Installation of the UNS-1E FMS. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement RFMS-1040 Issue 1 or later revisions of the above listed documents approved by EASA.Lees Certification Plan CP-1040 Issue 1 & MDL 1041-01 Issue 5. | This kit/STC can be installed on helicopter that has installed the kitDualBendix/King ED551A EHSI, EASA approved EASA.R.S.01445. | 14.09.2011 |
| 10036591 | LEES AVIONICS Ltd. | | | | SEE EASA APPROVED MODEL LIST | GNS430W/GNS530W Installation for BRNAV | CS-23 Amendment 2The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Multi Model List MML-1253 Issue 1, Dated 07 Sept 2011MasterData List MDL-1253 Issue 1, Dated 19 Sept 2011Change Statement CS-1253 Issue 1, Dated 02 Sept 2011Instructions for Continued Airworthiness ICA-1253 Issue1, Dated 02 Sept2011FlightManual Supp (...) | None | 30.09.2011 |
| 10039936 | LEES AVIONICS Ltd. | | | | BN2A-20, BN2A-21, BN2A-26, BN2A-27, BN2A-8, BN2A-9, BN2B-20, BN2B-21, BN2B-26, BN2B-27 | Garmin G600 PFD/ND. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval:CS23 Amendment 2The requirements for environmental protection and the associatedcertif (...) | Master Data List: MDL-1260 Issue 1 dated 31May12;Change Statement: CS-1260 Issue 1 dated 31May12;Flight Manual Supplement: AFMS-1260 Issue 1 or later revisions of the above listeddocuments approved by EASA. | None. | 04.06.2012 |
| 10043087 | LEES AVIONICS Ltd. | EASA.A.S.03637 | | | PA-42 | Upgrade of Mode S transponders for enhanced surveillance. | | -. Installation in accordance with Lees Avionics Ltd. Change numberHL/MOD/994-.,AFMS Lees Avionics Ltd. Supplement 994001, Issue 1, or later EASAapproved revision | none | 07.05.2009 |
| 10034642 | LETH & ASSOCIATES, LLC | | | FAA STC ST01857SE | 757-200 | Max Zero Fuel Weight Increase | The Certification Basis for the original product remains applicable tothis certificate.The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this approval. | Leth and Associates Drawing List OLO226, Revision D, 26th August 2009Leth and Associates AFM Supplement OLO224, Revision NC, 2nd December2010Leth and Associates Weight & Balance Supplement OLO225(-,-,->2ndDecember 2010or later revisions ofthe above (...) | Only applicable to a/c converted to Freighter by Precision ConversionsSTC (ref: EASA.IM.A.S.01423) | 18.04.2011 |
| 10035929 | LIBERTY AEROSPACE INC. | | | FAA STC SA03488AT | XL-2 | S-TEC system 30 autopilot installation on Liberty XL-2 series in accordance with the followingOption 1 (for use with S-Tec Model ST-901 GPS Steering Converter Unit);Chippewa Aerospace Master Data List (MDL) n° XL2-2210-01-0001, revision4, dated Nove (...) | The Certification Basis for the original product remains applicable tothis certificate/ approval.The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | AFM Supplement for S-TEC System 30 autopilot (DG Only) P/N 135A-970-530Initial releaseAFM Supplement for S-TEC System 30 autopilot integrated with an AspenEFD1000 PFD/MFD P/N 135A-970-531 Initial releasor later revisions of the above listed documen (...) | Option 1: Chippewa Aerospace Inc Document XL2-2210-01-0002 Rev IR, dated June 18, 2007(->->) or later FAA approved revision - Instructions forContinued Airworthiness (ICA).Option 2: Liberty Aerospace Document 135A-970-110, rev D, dated July 19,2010, (...) | 28.07.2011 |
| 10015453 | LIFEPORT, Inc. | EASA.IM.A.S.01198 | | FAA STC SA01093SE | CESSNA 525 CESSNA 525A | Installation and inspection of emergency medical stretcher and base, oxygen system in accordance with FAA STC SA01093SEor later EASA approved revision. | | - Approval Report, item 5;- Airplane Flight Manual Supplement EGR-689-11, Revision IR, dated May 31, 2002, or later approved revision. | | 13.02.2007 |
| 10015454 | LIFEPORT, Inc. | EASA.IM.A.S.01199 | | FAA STC ST00909SE | LEARJET 45 | Installation of emergency medical stretchers and bases, oxygen and electrical systems and associated Lifeport Inc. Report No. EGR-644-12 Revision IR dated November 6 2000 & MasterDrawing List EGR-644-00 revision R dated 25 April 2001 or later approved revision | | Lifeport Inc. Report No. EGR-644-12 Revision IR dated November 6 2000 & Master Drawing List EGR-644-00 revision U dated7 January 2005 or later EASA approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph 'Description'. Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 21.06.2005 |
| 10015551 | LIFEPORT, Inc. | EASA.IM.A.S.01445 | | FAA STC SA00273WI | BEECH 100 BEECH 200 BEECH 300 BEECH 65-90 BEECH 65-A90 BEECH B90 BEECH C90 BEECH C90 A BEECH E90 BEECH F90 | Installation of a Patient Loading and Utility System (PLUS) | | Lifeport Master Drawing List 520-900, revision C, dated 27 June 1994 or later approved revisionSupplement Flight ManualSupplement 35 for Lifeport PLUSSupplement No5, dated 28 December 1994 or later approved revision | Modification must be used with AFMS 35 for Lifeport PLUS | 19.09.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------|-------------------|----------------|----------------------|--|---|---|--|---|---------------|
| 10015614 | LIFEPORT, Inc. | EASA.IM.A.S.01553 | REV. 1 | FAA STC ST01780LA | CESSNA 560 XL | Installation of a Patient Loading and Utility System (PLUS)in accordance with FAA STC ST01780LA or later EASA approved revision>Note: Revision 1 issued to correct a type in the Original STC number. | | Master Drawing List, Nr. MDL-777 | Ref. AFM Supplement EGR-777-11, Revision A, dated June 24, 2005, or later FAA approved revision | 24.07.2009 |
| 10015646 | LIFEPORT, INC. | EASA.IM.A.S.01619 | | FAA STC SA00635SE | 208 208A 208B | Installation of a Patient Loading and Utility System (PLUS)in accordance with FAA STC SA00635SE or later EASA approvedrevision. | | See item 5 of approval report (attachment) | 1. Airplane Flight Manual supplement EGR-571-11, Revision A, dated January 11, 2002, or later FAA approved revision. | 16.02.2006 |
| 10016288 | LIFEPORT, Inc. | EASA.IM.A.S.02583 | | FAA STC SA02060LA | BEECH 1900D | Emergency Medical System | | FAA approved Master Drawing List, Doc. No. MDL-822-00, Rev.E, dated 09-Jul-2007, or later approved revisionFAA approved Airplane Flight Manual Supplement, Doc. No. EGR-822-11, Rev. A, dated 31-Jul-2007, or later approved revisionInstructions for Con (...) | The installer of this STC must ensure that, if the life vests have been removed with the seats, a life vest for the occupant of thelitter system is installed and available to the occupant. | 04.10.2007 |
| 10016483 | LIFEPORT, Inc. | EASA.IM.A.S.02837 | | FAA STC ST02119LA | CESSNA 750 CITATION X | Emergency Medical System | | Master Drawing List, MDL-833-00, Revision D Airplane Flight Manual EGR-833-11, Revision A Instructions for Continued Airworthiness, EGR-833-12-A | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 30.07.2008 |
| 10016489 | LIFEPORT, Inc. | EASA.IM.A.S.02845 | | FAA STC SA00096LA | BAE 125 SERIES 1000A BAE 125 SERIES 1000B BAE 125 SERIES 800A BAE 125 SERIES 800B HAWKER 1000 HAWKER 800 HAWKER 800 XP HAWKER HS 125-700A | Installation of a Patient Loading and Utility System. | | Installation in accordance with LifePort Inc. Master Drawing List as defined in AML No. EASA.IM.A.S.02845 to this STC. Dated 30 July 2008 or later approved revisions.Operation in accordance with LifePort Inc. Manual Supplement for LifePortPLUS Syste (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 30.07.2008 |
| 10016492 | LIFEPORT, Inc. | EASA.IM.A.S.02854 | | FAA STC ST00154SE | CESSNA 500 CESSNA 550 CESSNA 560 CESSNA S550 | Emergency Medical System | | Certification Plan EGR-700-10-A of 03 April 2002Instructions for Continued Airworthiness EGR-700-12 of 04 April 2002. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 27.08.2008 |
| 10016578 | LIFEPORT, Inc. | EASA.IM.A.S.02977 | REV. 1 | FAA STC SA00528SE | PA-31, PA-31-300, PA-31-350, PA-31P, PA-31T, PA-31T1, PA-31T2, PA-31T3, PA-31P-350, PA-42, PA-42-1000 | Emergency Medical System (The Emergency Medical System attaches only to the existing aircraft seat tracks. The system will be a free standing unit that can be rapidly installed and removed from the cabin). | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | FAA Approved Flight Manual Supplement AFMS for Life Port PLUS System, Supplement No.13, Rev. E, July 21 2004 - Master Drawing List for PLUS/Aero Sled Installation in the Piper PA-31and PA-42 Series Aircraft, Doc. Nr. EGR-556-00, FAA Approved 20. Ju (...) | None. | 30.04.2010 |
| 10016933 | LIFEPORT, Inc. | EASA.IM.R.S.01319 | | FAA STC SR01948LA | SIKORSKY S-92A | Installation of a triple litter stacking system | | Master Drawing List for the Triple Litter Stacking Installation, MDL-811-00, Revision D, dated December 14, 2006 or later approvedrevisionAirplane Flight Manual Supplement No. EGR-811-11, dated January 3, 2007, or later approved revision is required as part of this installation | The installation should not be incorporated in any aircraftunless it is determined that the interrelationship between this installation and any previously approved configurationswill not introduce any adverse effect upon the airworthiness of the air (...) | 02.05.2007 |
| 10016963 | LIFEPORT, Inc. | EASA.IM.R.S.01363 | | FAA STC SR02054LA | AGUSTA AB-139 AGUSTA AW-139 | Installation of an Emergency Medical System | | Master Drawing List MDL-831-100, Revision J dated June 12, 2007 or later FAA approved revisionApproved Rotorcraft Flight Manual Supplement for Emergency Medical System in the Agusta AB139 and AW139 Helicopter, document EGR-831-11 approved by FAA June 15, 2007 or later approved revision. | 1. The installer of this STC must be ensure that, if the life vests have been removed with the seats, a life vest for the occupant of the litter system is installed and available to the occupant. | 19.09.2007 |
| 10016978 | LIFEPORT, Inc. | EASA.IM.R.S.01381 | | FAA STC SR09082RC | BELL 222 BELL 230 | Air medical System. This STC has been classified as "BASIC" by the FAA and has been approved under the PTVP. | | 530-2220 revision Z dated 10 August 2007 or later approved revision drawing List;EGR-814-11 Revision A dated 12 September 2007 or later approved revision - Rotorcraft Flight Manual Supplement | | 05.03.2008 |
| 10016979 | LIFEPORT, Inc. | EASA.IM.R.S.01382 | | FAA STC SR00360DE | BELL 222 BELL 222B BELL 222U BELL 230 | Medical System. This STC has been classified as "BASIC" by the FAA and has been approved under the PTVP. | | EGR-515-00 revision S dated 10 August 2007 or later approved revision - Master Drawing List;EGR-515-12 revision A dated12 September 2007 or later approved revision - Flight Manual Supplement;EGR-515-15 revision B dated 2 July 2007 or later approved revisionInstruction for continued Airworthiness | | 05.03.2008 |
| 10030942 | LIFEPORT, Inc. | | | FAA STC ST01891LA-D | SEE EASA APPROVED MODEL LIST | Multi Models Commercial Medical Stretcher/validation of FAA STC ST01891LA-D | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approval.Certification basis as per TCDS except:-\$25,562 (when relevant) addressed per TGM/25/1 (...) | FAA approved Model List (AML) No. ST01891LA-D, dated May 27, 2010-FAA approved Airplane Flight Manual Supplement No. EGR-808-11, Rev Ddated May 27, 2010or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Deci (...) | This STC does not constitute an approval for installation. Compliancewith following requirements below need to be demonstrated through adesign approval for commercial medical stretcher installation in eachaircraft model per approved Model list § 25. (...) | 15.07.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|----------------------|--|--|---|---|--|---------------|
| 10032771 | LIFEPORT, Inc. | | | FAA STC ST02286LA | CESSNA 680 | Installation of a Medical Patient Loading and Utility System (PLUS) | EASA TCDS IM.A.033The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product ar (...) | Master Drawing List MDL-845-00, Revision T from 11 Nov. 2009Airplane Flight Manual Supplement EGR-845-11, Revision A from 22 Feb.2010Instructions for Continued Airworthiness EGR-845-12, Revision A from 25Jan. 2010or later revisions of the above list (...) | None | 30.11.2010 |
| 10015384 | LITEAIR AVIATION PRODUCTS Inc. | EASA.IM.A.S.01078 | | FAA STC ST01091SE | BOEING 737-100 BOEING 737-200 BOEING 737-200C BOEING 737-300 BOEING 737-400 BOEING 737-500 BOEING 737-600 BOEING 737-700 BOEING 737-700C BOEING 737-800 BOEING 737-900 | Installation of Lightweight Window Plugs. | | -FAA STC No ST01091SE.-LiteAir Aviation Products Inc MasterDrawing List LAP1004-D01 revision A dated July 9, 2003 or later FAA-approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.04.2005 |
| 10015385 | LITEAIR AVIATION PRODUCTS Inc. | EASA.IM.A.S.01079 | REV 1 | FAA STC ST00758SE | BOEING 767-200 SERIES BOEING 767-300 SERIES | Installation of Lightweight Window Plugs | | LiteAir Aviation Products Inc master Drawing List LAP1002-D01 revision IR dated October 22, 1999 or later FAA approved revision | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 26.03.2009 |
| 10015386 | LITEAIR AVIATION PRODUCTS Inc. | EASA.IM.A.S.01080 | | FAA STC ST00638SE | BOEING 747-100 BOEING 747-100B SUD BOEING 747-200B BOEING 747-200C BOEING 747-300 BOEING 747-400 BOEING 747-400D BOEING 747-SR | Installation of upper deck lightweight window plugs | | LiteAir Aviation Products Inc. Master Drawing List LAP1001-D02 revision IR dated June 13, 1998 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 18.11.2005 |
| 10015387 | LITEAIR AVIATION PRODUCTS Inc. | EASA.IM.A.S.01081 | | FAA STC ST00353SE | BOEING 747-100 BOEING 747-100B SUD BOEING 747-200B BOEING 747-200C BOEING 747-300 BOEING 747-400 BOEING 747-400D BOEING 747-SR | Installation of Main Deck Lightweight Window Plugs. | | -FAA STC No ST00353SE.-LiteAir Aviation Products Inc. Master Drawing List LAP1001-D01 revision D dated June 18, 1996 or later FAA-approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.04.2005 |
| 10016139 | LITEAIR AVIATION PRODUCTS Inc. | EASA.IM.A.S.02335 | | FAA STC ST01732SE | BOMBARDIER CL-600-2B19 | Installation of light weight window plug. | | FAA STC ST01732SE Lite Air Aviation Products MDL LAP1007-D01, Revision IR dated January 15., 2007 or later approved revision. | N/A | 15.03.2007 |
| 10016622 | LITEAIR AVIATION PRODUCTS Inc. | EASA.IM.A.S.03053 | | FAA STC ST01872SE | BOMBARDIER DHC-8-400 SERIES BOMBARDIER DHC-8-401 BOMBARDIER DHC-8-402 | Installation of lightweight window plug.Ref. FAA STC ST01872SE, dated 09 September 2008. | | 1. Installation in accordance with LiteAir Aviation Products Inc. Master Drawing List LAP1008-D01, Original Issue, dated 01 August 2008 *)2. Maintenance must be performed in accordance with Light Weight Window Plug Installation, Document LAP2 (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 26.11.2008 |
| 10032950 | LITEAIR AVIATION PRODUCTS Inc. | | | FAA STC ST02073SE | 777-200, -300, -200LR, -300ER | Lightweight window plug Installation | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval | FAA STC ST02073SE, dated July 26,2010LiteAir Certification Plan, LAP1009-R10, Revision A, dated March 11,2010LiteAir Master Drawing List, LAP 1009-D01, Revision A, dated June 23,2010LiteAir Installation Drawing, LAP2009, Revision B, dated June 23, 2 (...) | None | 09.12.2010 |
| 10042949 | LITEAIR AVIATION PRODUCTS Inc. | EASA.IM.A.S.01077 | | FAA STC ST01237SE | B757-200, B757-200CB, B757-300 | Installation of Lightweight Window Plugs. | | -FAA STC ST01237SE.-LiteAir Aviation Products Inc Master Drawing List LAP1005-D01 revisionIR dated May 01, 2003 or later FAA-approved revision. | This STC is approved only for the product configuration as defined inthe approved design data referred to in the paragraph "Description".Compatibility with other aircraft/engine configurations shall bedetermined by the installer. | 22.04.2005 |
| 10016048 | LIVETV | EASA.IM.A.S.02220 | | FAA STC ST10056SC | AIRBUS A320-232 | Installation Of Electronic Equipment Rack | | Master Data List, Document Number 2284-001 Revision N, dated 03rd March 2003 or later EASA approved revision.Instructions for Continued Airworthiness, Document Number CD-2002-002,Revision IR, dated 14th August 2002, or later EASA approvedrevision.Aircraft Flight Manual Supplement, Document Number | As per FAA STC ST10065SC the following STCs# must be installed concurrently with this STC:ST00788SE, ST02483AT and ST01447NY. | 29.05.2007 |
| 10016049 | LIVETV | EASA.IM.A.S.02221 | | FAA STC ST02483AT | AIRBUS A320 SERIES | Installation Of a Cabin Mounted Video Camera System | | Master Drawing List, Document Number LVTY-002-MDL Revision D, dated 07th April 2003, or later EASA approved revision.Instructions for Continued Airworthiness, Document Number LVTY-002-CAW, Revision B, dated 22nd February 2002, or later EASA approved (...) | As per FAA STC ST10065SC the following STCs# must be installed concurrently with this STC:ST10065SC, ST00788SE, and ST01447NY. | 29.05.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|-------------------------------|---|---|---|--|---|---------------|
| 10016050 | LIVETV | EASA.IM.A.S.02222 | | FAA STC ST00788SE | AIRBUS A320-200 | Installation of a Live TV Satellite Television System | | Master Drawing List, Document Number LTV933-D01-A320, Revision J, dated 24th May 2005 or later EASA approved revision. | As per FAA STC ST10065SC the following STC#s must be installed concurrently with this STC: ST10065SC, ST02483AT and ST01447NY. | 29.05.2007 |
| 10016051 | LIVETV | EASA.IM.A.S.02223 | | FAA STC ST01447NY | AIRBUS A320-200 | Installation Of Harris Ground Wireless Aircraft Data Link & Partial Power Provisions for Installation of a Cabin Surveillance System | | Master Drawing List, Document Number 3092086-MDL, Revision C, dated 29th July 2004, or later EASA approved revision. Instructions for Continued Airworthiness, Document Number 3092085, Revision B, dated 31st January 2002, or later EASA approved revision. | As per FAA STC ST10065SC the following STC#s must be installed concurrently with this STC: ST10065SC, ST02483AT and ST00788SE. | 29.05.2007 |
| 10016503 | LIVETV | EASA.IM.A.S.02867 | | FAA STC ST03177AT | EMBRAER ERJ 190-100 ECJ | FAA STC ST03177AT: Installation of LiveTV LTV 2000 In-Flight Entertainment System | | -MDL J1901TR0010-MDL, Revision E, dated June 14, 2006 *) -AFM Supplement J1901TR0010-AFM dtd. October 18, 2005 *) -Instructions for Continued Airworthiness J1901TR0010-ICA Rev. IR dtd August 3, 2005 *) or later approved revision law EASA ED Decision 2007/001/C or later revisions of this Decision | Installation based on EMBRAER provisions | 05.09.2008 |
| 10041796 | LLOPIS BALLOONS | | | | 1. BASKET: THUNDER & COLT 40+B 2. BASKETS: C5, C5L, C7, C8, C8L, C9 and C11 baskets and Cameron CB8202 "Thunder & Colt 40+B" baskets 16-guests kit applicable on Ultramagic C12 basket | Modifications "Diners a bord" 2-guests kit applicable on Ultramagic C5, C5L, C7, C8, C8L, C9 and C11 baskets and Cameron CB8202 "Thunder & Colt 40+B" baskets 16-guests kit applicable on Ultramagic C12 basket | The Certification Basis for the original product remains applicable to this certificate/ approval | Supplement FM Llopis Balloons 01-2012 or later revisions of the above listed documents approved by EASA. Master Drawing List MDL-01-2012 | Modification to Thunder & Colt baskets limited to S/N 943 and 1834 when any guests kits are in place, the supplemental light manual "supplément au manuel de vol Llopis Balloons pour modification en configuration diners a bord" must be used. | 22.10.2012 |
| 10016038 | LOPRESTI SPEED MERCHANTS Inc. | EASA.IM.A.S.02207 | REV. 2 | ST02893AT | BAE 125 SERIES 1000A BAE 125 SERIES 800A HAWKER 750 HAWKER 800 HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of a Boom Beam Taxi/Landing Light System. HID Wing Taxi/Landing Light System, HID Nose Taxi Light System | | LoPresti Master Drawing List (MDL), Report No. 47, Rev. A or later approved revisions | | 16.11.2007 |
| 10016161 | LP AERO PLASTICS, INC. | EASA.IM.A.S.02383 | | FAA STC SA00204NY | BEECH 35 BEECH 35-33 BEECH 35-A33 BEECH 35-B33 BEECH 35-C33 BEECH 95 BEECH 95-35 BEECH 95-A45 BEECH 95-B55 BEECH A35 BEECH B95 BEECH B95A BEECH C35 BEECH E35 BEECH F35 BEECH G35 BEECH H35 BEECH J35 BEECH K35 BEECH M35 BEECH N35 BEECH P35 BEECH S35 | Installation of 250#, 312# or 375# thick one piece windshields (two piece windshield to one piece wind shield conversion). | | LP Plastic Aero Plastics, Inc., Single Piece Extra Thick Windshield Installation Drawing List: Report No. Z71/CON dated April 27, 1994. | 1. This approval should not be incorporated in any aircraft models listed in this STC on which other approved modifications are incorporated, unless it is determined that the interrelationship between this change and any of those previously incor (...) | 20.04.2007 |
| 10015134 | LTB SAMMET GmbH | EASA.A.S.03487 | | EASA STC EASA.A.S.C.01008 AND | CESSNA 150D CESSNA 150E CESSNA 150F CESSNA 150G CESSNA 150H CESSNA 150J CESSNA 150L CESSNA 150M CESSNA A150L CESSNA F150-G CESSNA F150-H CESSNA F150-J CESSNA F150-K CESSNA F150-L CESSNA F150-M CESSNA FA-150-K | Installation of engine Rotax 912 F3 and installation of Propeller Hoffmann HO-V352F/170FQ on Cessna Models 150H, F150H, A150K, FA150K, A150L, FA150L, 150K, F150K, 150L and F150L only. Installation of engine Rotax 912 S3 (100PS) and installation of Pro (...) | | Aircraft installed with Rotax 912 F3; Cessna 150H, F150H - Flight Manual Operation Instruction Hoffmann No. 605 Issue 09.98, LBA approved October 1998 or later approved revision. Cessna A150K, FA150K, A150L, FA150L - Flight Manual and Operation Instru (...) | 1. Spin is prohibited to all models. 2. Aerobac including Spin is prohibited to Cessna A150K, FA150K, A150L and FA150L. 3. Prior to installation of this modification the installer must determine that the interrelationship between this modification and (...) | 25.02.2008 |
| 10013924 | LUFTHANSA TECHNIK AG | EASA.A.S.02836 | REV. 4 | | AIRBUS A318-112 | Structural Floor Panel Reconfiguration The limitation for divans that they must not be occupied for taxi, take off and landing has been suspended for 1315 FH or 14 months whatever occurs first, starting at June 26th, 2008. The modification is described (...) | | Supplemental Type Definition Document, ref. Doc. No. STD00-06/09M, issue 08, dated 17. December 2008, or later approved revision. Aeroplane Flight Manual Supplement A-00-07/04-FM revision 7, dated 09. October 2008, or later approved revision. Instruct (...) | 1. Side facing divans may not be used for taxi, take off and landing after 1315 FH or 14 months whatever occurs first, starting at June 26th, 2008. 2. Prior to installation of this modification the installer must determine that the interrelationship (...) | 18.12.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|----------------|----------------|----------------------|--|---|---------------------|--|--|---------------|
| 10013943 | LUFTHANSA TECHNIK AG | EASA.A.S.00382 | REV 1 | | AIRBUS A340-300 | Installation of Antenna and Receiver Transmit System and Network Services(acc.: STDD 00-03/11M Issue 02) | | AFM Supplement: A-00-04/072-AFMICA: R-00-04/023 | 1. The IEEE 802.11 a, b, g (as customized) Wireless Local Area Network (WLAN) introduced by this modification, may be activated onboard Airbus A340-600 Aircraft for operation with commercial IEEE 802.11a, b, g & h compliant WLAN enabled portable el (...) | 15.11.2005 |
| 10013964 | LUFTHANSA TECHNIK AG | EASA.A.S.00654 | REV. 2 | | A340 | Installation of non-electrical Floor Path Marking System Guideline® | | - STC TA0018- Supplemental Type Definition Document STDD33-97/01M, Issue 07 dated 09.11.2006- Applicant's Declaration of Compliance dated 15.11.2006- Airplane Flight Manual Supplement A-33-06/367-AFM-Instructions for Continued Airworthiness or later r (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 14.09.2012 |
| 10013967 | LUFTHANSA TECHNIK AG | EASA.A.S.00701 | REV 1 | | BOEING 747-300-400 | Radome Installation for Live TV Antenna Installation | | - Lufthansa Technik Supplemental Type Definition Document, Doc. No STDD 53-04/01M, Issue 01, rev. 00 dated April 5, 2005 or later approved Revision (for B747-400)- Lufthansa Technik Supplemental Type Definition Document, Doc. No STDD53-08/04M, (...) | 1. Maintenance requirements, Doc. No. C-AAL900-MR shall be incorporated in the aircraft maintenance documentation.2. This STC is approved only for the product configuration as defined in the approved design data. Compatibility with other aircraft (...) | 23.04.2009 |
| 10013977 | LUFTHANSA TECHNIK AG | EASA.A.S.01000 | | | BOEING 767-300 | Installation of an Optical Quick Access Recorder (OQAR) according to LHT Engineering Bulletin No. 767-EB31-0203, dated 09.09.04 or later approved Revision for FOQA Purpose | | | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other Aircraft configurations shall be determined by the installer. | 17.08.2005 |
| 10013978 | LUFTHANSA TECHNIK AG | EASA.A.S.01001 | | | BOEING 737-300 | Installation of an Optical Quick Access Recorder (OQAR) according to LHT Engineering Bulletin No.737-EB31-0202, dated 29.09.04 or later approved Revision for FOQA Purpose. | | | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other Aircraft configurations shall be determined by the installer. | 17.08.2005 |
| 10014013 | LUFTHANSA TECHNIK AG | EASA.A.S.01056 | REV. 1 | | A318/A319/A320/A321 | Installation of non-electrical Floor Path Marking System Guideline® | | Supplemental Type Definition Document STDD33-04/03M, issue 02 dated 28/11/2005 Compliance Record List CRL33-04/03M, issue 02 dated 20/01/2006 Airplane Flight Manual Supplement A-33-04/4621-AFMA Applicant's Declaration of Compliance dated 03/02/2006 or later (...) | | 14.09.2012 |
| 10014043 | LUFTHANSA TECHNIK AG | EASA.A.S.01121 | | | AIRBUS A330-200 | A330-200 Air Algeria Quick Change VIP Kit OnePlus. | | -Supplemental Type Definition Document: STDD00-04/20M, Issue 1.-Compliance Record List: CRL00-04/20M, Issue 1.- Airplane Flight Manual Supplement acc.: A-00-05/084-AFM Rev. 03.-Instructions for Continued Airworthiness.-Master Minimum Equipment List (MMEL) A-00-05/084-MMEL | | 31.03.2005 |
| 10014079 | LUFTHANSA TECHNIK AG | EASA.A.S.01171 | | | BOEING 777-300 | Installation of PC Power and PC Data Outlets on the existing B/C Seats and installation of new ECO Class Seats with new P/N and PC-Power/PC-Data Outlets on Boeing 777-300 A/C of Cathay Pacific Airways. | | -Supplemental Type Definition Document No STDD25-03/09M, Issue 01/ Rev. 1 or later approved Revisions. | | 01.03.2005 |
| 10014126 | LUFTHANSA TECHNIK AG | EASA.A.S.01236 | | | BOEING 747-284B BOEING 747-SP | Flight Management System (FMS), Activation of Advisory VNAV Modes according to Lufthansa Technik Supplemental Type Definition Document STDD34-05/02M, Issue 01, Rev 00, dated April 20, 2005 or later approved revision | | Airplane Flight Manual supplement A-34-05/221-AFM, revision IR, dated April 06/05 or later approved revision Instructions for Continued Airworthiness Doc. No. CS34-0381-01, dated 05.04.2005 | | 30.05.2005 |
| 10014158 | LUFTHANSA TECHNIK AG | EASA.A.S.01286 | | | BOEING 767-300 | B-767- Cabin Layout Change - Air Europe | | Supplemental Type Definition Document Doc. No. STDD 00-05/03M, Issue 1. Instructions for Continued Airworthiness Doc. No. ICA 25-1009-00 dated May 30, 2005 or later approved issue | | 21.06.2005 |
| 10014191 | LUFTHANSA TECHNIK AG | EASA.A.S.01333 | REV 1 | | BOEING 767 SERIES | Installation of Guideline-Self Illuminating Floor Path Marking System. | | STDD No. 33-05/02M, Issue 2 AFMS Doc. No. A-33-05/314-AFM Supplement for Boeing 767 Series LHT Doc. No. A-33-05/331-AFM, dated October 31/05 or later approved revision | | 14.12.2005 |
| 10014193 | LUFTHANSA TECHNIK AG | EASA.A.S.01335 | | | BOEING MD 11 BOEING MD 11F | Navigation - ATC system, Installation of Upgrade for Enhanced Surveillance and associated Airplane Flight Manual Supplement. | | LHT Engineering Bulletin Ref. MD11-EB34-0357, Rev. IR dd. 14 January 2005, - LHT Airplane Flight Manual Supplement doc. No. A-34-04/6066-AFM, Rev. IR dd. 6 June 2005 | | 29.06.2005 |
| 10014196 | LUFTHANSA TECHNIK AG | EASA.A.S.01338 | | | BOEING 737-300 BOEING 737-400 BOEING 737-500 | Navigation - Introduction of ATC Mode S Enhanced Surveillance | | -Supplemental Type Definition Document STDD34-05/05M, Issue 01 Rev 0 or later approved revision-Airplane Flight Manual Supplement Doc No A-34-05/261-AFM Rev IR dated June 27, 2005 or later approved revision-Maintenance Manual Supplement Doc No DR34- | | 26.07.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|----------------|----------------|----------------------|--|--|---|--|--|---------------|
| 10014216 | LUFTHANSA TECHNIK AG | EASA.A.S.01377 | | | BOEING 737-300 | *Avionics Upgrade by- Installation of TCAS-Installation of Dual Mode S Transponder-Installation of Add. Parameters for Digital FDR- Digital Flight Data Acquisition Unit, Software Upgrade- Installation of Enhanced Ground Proximity Warning System* | | - Supplemental Typ Definition Document STDD34-05/04M Issue1, Rev. 00 dated Aug 19, 2005 or later approved Revision.- Airplane Flight Manual Supplement Doc. No. A-00-05/322 AFM dated Aug 17, 2005 or later approved Revision.- Instructions for Continio (...) | | 01.09.2005 |
| 10014218 | LUFTHANSA TECHNIK AG | EASA.A.S.01379 | REV. 1 | | AIRBUS A340-600 | Installation of Antenna, Receiver Transmit System, and Network services.This STC installs through the cabin a set of equipment to allow dataexchange.The Revision 1 has introduced some modifications through the initial STC design by upgrading the exis (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Supplemental Type Definition Document, ref. STDD46-11/14, Issue 1; Instruction for Continued Airworthiness, ref. R-00-04/120, Revision 5; Aircraft Flight Manual Supplement, ref. A-00-04/133-AFM, Revision 2; or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 27.02.2013 |
| 10014225 | LUFTHANSA TECHNIK AG | EASA.A.S.01388 | | | AIRBUS A320 AIRBUS A321 | ATC, System Upgrade for Enhanced Surveillance, acc.: STDD 34-05/06M | | AFM Supplement: A-34-05/260-AFMICA: DR34-0379-02 | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.07.2005 |
| 10014227 | LUFTHANSA TECHNIK AG | EASA.A.S.01390 | | | AIRBUS A330-300 | Integration of Lower Deck Mobile Crew Rest | | Supplemental Type Definition Document STDD00-04/11M Issue 01 Compliance Record List CRL00-04/11 Issue 1 Instructions for Continued Airworthiness | | 14.07.2005 |
| 10014228 | LUFTHANSA TECHNIK AG | EASA.A.S.01391 | | | AIRBUS A340-300 | Integration of Lower Deck Mobile Crew Rest | | Supplemental Type Definition Document STDD00-04/15M Issue 01 Compliance Record List CRL00-04/15 Issue 1 Instructions for Continued Airworthiness | | 14.07.2005 |
| 10014246 | LUFTHANSA TECHNIK AG | EASA.A.S.01456 | | | BOEING 767-300 | ATC, System Upgrade for EHS | | Supplemental Type Definition Document in STDD34-04/03M Issue 01, Rev 01 or later approved revision/Airplane Flight Manual Supplement A-34-04/6070-AFM Rev IR or later approved revision | Certification Basis of the Change: CS 25 Change 0, additionally JAA Leaflet No 18 (Mode-S EHS) | 10.04.2006 |
| 10014256 | LUFTHANSA TECHNIK AG | EASA.A.S.01483 | REV. 1 | | FOKKER F.28 MARK 0070 FOKKER F.28 MARK 0100 | Installation of non-electrical Floor Path Marking System Guideline® | | 1. Supplemental Type Definition Document STDD33-05/01M, issue 01 dated 03.03.20062. Airplane Flight Manual Supplement A-33-05/328-AFM3. Applicant's Declaration of Compliance dated 17.03.20063 or later revisions of the above listed documents approved by EASA | | 14.09.2012 |
| 10014257 | LUFTHANSA TECHNIK AG | EASA.A.S.01486 | REV 1 | | AIRBUS A330-200 AIRBUS A330-300 | Structural mounting provision for large antenna and random for Live TV Tailwind 560 | | Airplane Flight Manual Supplement: A-00-06/136-AFM Rev. IR, or later approved revision. Supplemental Type Definition Document Ref. STDD 53-05/02M, Issue 02 / Rev. 00 dated 7 December 2006. Instructions for Continued Airworthiness Ref. R-000-06-0149, Rev. 03 dated 7 December 2006 | 1. EASA CS25, without Amendment, Certification Specifications for Large Aeroplanes plus FAA Generic Issue Paper A-X "Structural Certification Criteria for Large Antenna Installations". 2. This STC is approved only for the product configuration as def (...) | 08.12.2006 |
| 10014258 | LUFTHANSA TECHNIK AG | EASA.A.S.01487 | | | AIRBUS A330-200 AIRBUS A330-300 | Communication - Modification/Upgrade of the In-Flight Entertainment System, Installation of Tailwing 560 Live TV Equipment. | | Airplane Flight Manual Supplement: A-00-06/136-AFM Rev. IR, or later approved revision. Instructions for Continued Airworthiness: R-00-06-0149 Rev. IR, or later approved revision. STDD No. "STDD 23-05/05M" | EASA CS25, without Amendment, Certification Specifications for Large Aeroplanes plus JAA INT/POL/25/3 Issue 2, JAA INT/POL/25/4 Issue 3, JAA Leaflet No. 17. | 07.04.2006 |
| 10014260 | LUFTHANSA TECHNIK AG | EASA.A.S.01488 | REV 1 | | BOEING 777-200 | Rearrangement of VIP cabin interior (Majlis) & systems and modification of pax entertainment system (IFE) including installation of tailwind 560 Live TV system. | | STC Definition Document - STDD00-05/06M issue 01 dated 11 June 2008 (or later approved issue) Compliance Record List CRL00-05/06M issue 1 Instructions for Continued Airworthiness # A-00-05/342-ICA rev IR (or later approved issue) EASA CRIs A-01 Iss 4 Certification Basis VIP Interior. | 1. This modification is limited to a/c serial number MSN29271.2. Occupancy of this zone is limited to Zero Passengers for Taxi, Take-off and Landing, pending substantiation of evacuation through the adjacent compartments | 12.06.2008 |
| 10014261 | LUFTHANSA TECHNIK AG | EASA.A.S.01489 | | | BOEING 777-200 | Introduction of Tailwind 550 Antenna and Live TV for VIP B777-200. | | STDD53-05/01M issue 01 dated 22 June 2006 AFM Supplement # A-00-05/342-AFM Instructions for Continued Airworthiness CSS3-0223-2 | This modification is limited to a/c serial number MSN29271. | 29.06.2006 |
| 10014262 | LUFTHANSA TECHNIK AG | EASA.A.S.01490 | | | AIRBUS A300 | ATC, System Upgrade for EHS | | Supplemental Type Definition Document STDD34-05/07M, Issue 1, Rev. 00 or later approved Revision/Airplane Flight Manual Supplement A-34-05/323-AFM Rev. IR, dated Jul 22/05 or later approved Revision | Certification Basis of the Change: CS 25, additionally JAA Leaflet No. 18 (Mod-S EHS) | 13.03.2006 |
| 10014269 | LUFTHANSA TECHNIK AG | EASA.A.S.01498 | | | BOEING 737-700 IGW | Cabin and IFE Modification as defined in Supplemental Type Definition Document STDD00 - 05/05 M, Issue 01, Rev.00 | | Lufthansa Technik Supplemental Type Definition Doc. No. STDD00 - 05/05 M, Issue 01, Rev. 00 or later approved Revision | | 25.10.2005 |
| 10014281 | LUFTHANSA TECHNIK AG | EASA.A.S.01518 | | | BOEING 737-600 SERIES BOEING 737-700 SERIES BOEING 737-800 SERIES BOEING 737-900 SERIES | Installation of Guideline-non-electrical Floor Path Marking System. | | Supplemental Type Definition STDD 33-05/03M/Airplane Flight Manual Supplement LHT Doc. No. A-33-05/324-AFM | | 25.10.2005 |
| 10014282 | LUFTHANSA TECHNIK AG | EASA.A.S.01519 | | | BOEING 747-400 | VVIP Cabin Installation STDD 25-05 / 06M issue 1, STDD 25-05 / 19 issue 1 | | ADoc: HAM WL A9-UAE, dated Oct 13, 2006 AFMS: A-00-06 / 181-AFM issue 1 Rev. 5/FA: A-00-06 / 137 | -- | 19.10.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|----------------|----------------|----------------------|------------------------------------|---|--|--|---|---------------|
| 10014283 | LUFTHANSA TECHNIK AG | EASA.A.S.01520 | | | BOEING 757-300 | Installation of Zonal Drying System | | Supplemental Type Definition Document STDD21-05/02M, Issue 1, Rev. 0, or later approved revision/Instructions for Continued Airworthiness CS21-0142-02, Rev. IR, or later approved revision | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 06.11.2007 |
| 10014292 | LUFTHANSA TECHNIK AG | EASA.A.S.01534 | REV. 1 | | B737-700 IGW | Revision of EASA.A.S.01534 IR per Major Change 0010002739-001: Operation at 8.99 psi maximum cabin differential pressure (Lower Cabin Altitude, LCA) Original STC Title: B737-700 IGW VIP Cabin 19 Pax. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: -CRI-D01 (Doors between Passenger Compartments) -CRI D-02 (Isolated Compartments) -CRI F (...) | Supplemental Type Definition Doc: STDD00-05/07M Issue 03, dtd Oct 22, 2012; - AFM Supplement Doc A-00-06/100-AFM, Revision 03, dated Jul 18/06; - Instructions for Continued Airworthiness (ICA) Doc A-00-06/100-ICA Issue 01, dated June 2, 2010; or later r (...) | Once LCA is activated (installation of FAA STC ST01697SE as validated per EASA STC 10042295) on an individual aircraft, the maintenance requirements defined in Doc A-00-06/100-ICA will apply for the life of this aircraft. This limitation remains valid a (...) | 03.12.2012 |
| 10014293 | LUFTHANSA TECHNIK AG | EASA.A.S.01535 | REV 1 | | BOEING 747-SP | ATC System, Upgrade to Enhanced Surveillance Note: This Revision expands the application range of the existing STC. | | Supplemental Type Definition Document No. STDD34-05/08, M, Issue 2, Rev. 00, dated 15 January 2007 or later approved revision; Airplane Flight Manual Supplement A-34-06/499-AFM, Revision IR, dated 22 Nov 2006 or later approved revision | 1. Certification Basis of the change: CS 25 Change O, additionally JAA Leaflet No. 18 (Mode-S EHS)2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compa (...) | 15.01.2007 |
| 10014308 | LUFTHANSA TECHNIK AG | EASA.A.S.01553 | | | AIRBUS A319 | Installation of new B/E Aerospace convertible and standard seats and cabin refurbishment to PG-design on Lufthansa's A319 aircrafts | | Certification Program CP25-05/20M, issue 03, dated 18 Jan 2006 Supplemental Type Definition Document STDD25-05/20M, issue 01 dated 30/01/2006 Compliance Record List CRL25-05/20M, issue 01 dated 24/01/2006 Applicant's Declaration of Compliance dated 03/02 | | 16.02.2006 |
| 10014309 | LUFTHANSA TECHNIK AG | EASA.A.S.01554 | | | BOEING 747-400 | B747-400 Guggenheim Seat Arrangement Modification from 384 passengers (main deck 16 FC, 28 BC, 310 Y) to 398 passengers (main deck 12 FC, 0 BC, 46 Y + 310 Y) according to STD00-05/10M | | Interior Arrangement according Design Document "Engineering Bulletin" Doc. No. 747-EB25-1068 | | 24.10.2005 |
| 10014316 | LUFTHANSA TECHNIK AG | EASA.A.S.01564 | | | AIRBUS A300-600 | *Installation of Communication System IFE Update'acc.: STDD23-05/07M | | AFM Supplement: A-24-05/6080-AFMICA: CS 53-0241-02 | Limited to s/n 354 | 10.11.2005 |
| 10014325 | LUFTHANSA TECHNIK AG | EASA.A.S.01578 | REV. 2 | | B747-400, B747-400F | Installation of non-electrical Floor Path Marking System Guideline® | | Supplemental Type definition Document STDD33-02/05M, issue 03 dated 21/03/2006 Airplane Flight Manual Supplement A-33-05/340-AFM Applicant's Declaration of Compliance dated 30/03/2006 or later revisions of the above listed documents approved by EASA | | 14.09.2012 |
| 10014326 | LUFTHANSA TECHNIK AG | EASA.A.S.01579 | | | B737-800 | Business Class Installation with Cirma Seats | | Supplemental Type Definition Document Ref. STDD 25-05/31M, Issue 2 dated 16 December 2005 | | 13.01.2006 |
| 10014347 | LUFTHANSA TECHNIK AG | EASA.A.S.01611 | | | AIRBUS A321 | Installation of new B/E Aerospace convertible and standard seats and cabin refurbishment to PG-design on Lufthansa's A321 aircrafts | | 1. Certification Program CP25-05/22M, issue 02, dated 20.01.2006. Supplemental Type Definition Document STDD25-05/22M, issue 01 dated 17.02.2006. Compliance Record List CRL25-05/22M, issue 01 dated 17.02.2006. Applicant's Declaration of Compliance dated 21.02.2006 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 03.03.2006 |
| 10014354 | LUFTHANSA TECHNIK AG | EASA.A.S.01619 | REV 4 | | BOMBARDIER CL-600-2B19 | CL-600-2B19 - Cabin Completion - 15 PAX # Configuration D. | | 1. Supplement Type Definition Document STDD00-05/12M, issue 04, dated November 13, 2007, or later approved revision. 2. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 3. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 4. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 5. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 6. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 7. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 8. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 9. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 10. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 11. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 12. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 13. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 14. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 15. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 16. 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Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 73. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 74. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 75. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 76. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 77. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 78. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 79. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 80. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 81. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 82. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 83. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 84. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 85. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 86. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 87. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 88. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 89. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 90. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 91. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 92. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 93. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 94. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 95. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 96. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 97. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 98. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 99. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. 100. Configuration Identification Document CID-15-01619-LHT, issue 03, dated November 2, 2007, or later approved revision. | 1. Certification Basis of Change: JAR-25 Change 13, additionally Configuration A and B-CRI D-01, Side facing Divan (Deviation) CRI F-01, Executive Power Systems (Guidance Material) JAA TGL No. 17, Passenger service and In-Flight Entertainment (IFE (...)) | 20.11.2007 |
| 10014359 | LUFTHANSA TECHNIK AG | EASA.A.S.01624 | | | BOEING 737-300 SERIES | Installation of new Recaro Seating convertible and standard seats and cabin refurbishment to PG-design on Lufthansa's Boeing B737-300 aircraft | | *Certification Program CP25-05/23M, issue 01, dated 01.02.2006 Supplemental Type Definition Document STDD25-05/23M Iss.01, Rev. 00, 27.02.06 | Certification Basis of the change: FAR25 as defined in FAA TCDS A16WE. The applicant showed compliance to CS 25.0785(h)(2), CS 25.811(d), (g), CS 25.812#, CS 25.0853(a), (c), (d), (f). | 13.03.2006 |
| 10014367 | LUFTHANSA TECHNIK AG | EASA.A.S.01637 | | | AIRBUS A319 | A319 CJ VIP Cabin 34 Pax | | Supplemental Type Definition Document STDD00-05/08M, issue 02 dated 18.07.2006 Airplane Flight Manual Supplement A-00-05/343-AFMS Applicant's Declaration of Compliance dated 19.07.2006 | 1. JAR-25 Change 112, FAR 25.981 (a) (b) Amend. 102 'Fuel Tank (ignition prevention)' 3. CRI D-03 'Type III Emergency Exit Access' dated 31. March 2006, CRI F-01 'Domestic Power' dated 18 May 2006. TGL No. 17 'Passenger Service and In-Flight Entert (...) | 21.07.2006 |
| 10014373 | LUFTHANSA TECHNIK AG | EASA.A.S.01647 | REV. 1 | | B717-200 DC-9, MD-30, MD-88, MD-90 | Installation of non-electrical Floor Path Marking System Guideline® | | Supplemental Type Definition Document STDD33-05/05M, issue 02 dated 27/03/2006 Airplane Flight Manual Supplement A-33-05/326-AFM Applicant's Declaration of Compliance dated 28/03/2006 or later revisions of the above listed documents approved by EASA | | 14.09.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|----------------|----------------|----------------------|---|---|---------------------|--|--|---------------|
| 10014395 | LUFTHANSA TECHNIK AG | EASA.A.S.01689 | | | AIRBUS A319 AIRBUS A320 AIRBUS A321 | Introduction of Link 2000+ | | Supplement Type Definition Document No. STDD23-05/08M, Issue 01, Rev. 00, dated 20 February 2007 or later approved revision Airplane Flight Manual Supplement Doc. No. A-23-06/357-AFM, revision IR, dated 29 January 2007 or later approved revision | Certification Basis of the Change: CS 25 Amdt 02 | 26.02.2007 |
| 10014398 | LUFTHANSA TECHNIK AG | EASA.A.S.01692 | | | BOEING 747-400 | B747-400 Bahrain Royal Flight, VIP Cabin Installation | | Supplemental Type Definition Document Doc No STDD 00-03/17M, Issue 01 or later approved Issue Airplane Flight Manual Supplement Doc No A-00-05/319-AFM, Rev 02 or later approved revision Instructions for Continued Airworthiness Doc No ICA-0-03/17 | | 16.12.2005 |
| 10014404 | LUFTHANSA TECHNIK AG | EASA.A.S.01699 | | | BOEING 737-500 | Installation of new Recaro Aircraft Seating convertible and standard seats and cabin refurbishment to PG-design on Lufthansa's Boeing B737-500 aircraft. | | Certification Programm CP25-05/25M, issue 01, dated 09.03.2006 Supplement Type Definition Document STDD25-05/25M Iss. 1, 18.04.2006 Compliance Record List CRL25-05/25M Iss. 1, 10.04.2006 Applicant's Declaration of Compliance ADOC dated 20.04.2006 | Certification Basis of the change: FAR25 as defined in FAA TCDS A16WE. | 23.05.2006 |
| 10014415 | LUFTHANSA TECHNIK AG | EASA.A.S.01715 | | | AIRBUS A320 | Installation of new B/E Aerospace convertible and standard seats and cabin refurbishment to PG-design on Lufthansa's A320 aircrafts | | 11. Certification Program CP25-05/21M, issue 02, dated 20.01.20062. Supplemental Type Definition Document STDD25-05/21M, issue 01 dated 07.02.20063. Compliance Record List CRL25-05/21M, issue 01 dated 07.02.20064. Applicant's Declaration of Compliance dated 14.02.2006 | | 17.02.2006 |
| 10014449 | LUFTHANSA TECHNIK AG | EASA.A.S.02502 | | | A320-200 | A320 Pax Capacity Increase from 156 to 168 Pax | | Supplement Type Definition Document STDD25-06/01M, issue 01 dated 25.05.06 Applicant's Declaration of Compliance dated 26.05.2006 | | 29.05.2006 |
| 10014472 | LUFTHANSA TECHNIK AG | EASA.A.S.02533 | REV 1 | | AIRBUS A340-300 | Installation of CAIR System in A340-300 | | - Supplemental Type Definition Document, Ref. STDD21-05/01M, Issue 2 dated 18/08/2006- Instructions for Continued Airworthiness, Ref. ICA-21-0150, Rev. 01 dated 01/08/2006 | | 18.08.2006 |
| 10014494 | LUFTHANSA TECHNIK AG | EASA.A.S.02556 | | | BOMBARDIER CL-600-2B19 | CL-600-2B19 - Cabin Completion - 16 PAX | | Supplemental Type Definition Document STDD00-06/06M, Issue 01, dated March 27, 2007 or later approved revisions Aeroplane Flight Manual Supplement A-00-06/461-AFM, Issue 01, dated January 05, 2007 or later approved revisions Instructions for Continued Airworthiness (I) (..) | 1. Certification Basis of the Change: JAR-25 Change 13, additionally: CRI F-01, Executive Power System (Guidance Material)CRI D-01, Side Facing Divans (Deviation)CRI D-04, Door between passenger compartments with integrated blocking functionality (D (...)) | 29.03.2007 |
| 10014504 | LUFTHANSA TECHNIK AG | EASA.A.S.02574 | | | BOEING 747 | Installation of an ATC Mode S - Enhanced Surveillance Upgrade in Boeing 747 series aircraft | | The modification is described in Supplemental Type Definition Document Navigation ATC Mode S EHS Upgrade, Lufthansa Technik Document STDD34-06/02M, issue 01, Rev. 00 Data package containing the following documents, and additional supporting data: 1. (...) | EASA approved Airplane Flight Manual Supplement, Lufthansa Technik Document A-34-06/135-AFM, Rev IR or later EASA approved revision is required part of this STC | 04.08.2006 |
| 10014507 | LUFTHANSA TECHNIK AG | EASA.A.S.02580 | REV. 1 | | A330 SERIES | Installation of non-electrical Floor Path Marking System Guideline® | | 1. Supplemental Type Definition Document STDD33-03/01M, issue 03 dated 12.01.20062. Airplane Flight Manual Supplement A-33-05/348-AFM3. Applicant's Declaration of Compliance dated 27.02.2006 or later revisions of the above listed documents approved by EASA | | 14.09.2012 |
| 10014509 | LUFTHANSA TECHNIK AG | EASA.A.S.02583 | | | AIRBUS A300-600 AIRBUS A310-300 | Installation of a new Stretcher for Kuwait Airways. | | Supplemental Type Definition Document STDD25-06/06M, Issue 1, Revision 1, dated 19 September 2007, (or any later approved Revision of this document) | | 15.11.2007 |
| 10014510 | LUFTHANSA TECHNIK AG | EASA.A.S.02585 | | | AIRBUS A320 | Installation of a new Stretcher for Kuwait Airways | | 1. Supplemental Type Definition Document, ref. Doc. No. STDD25-06/07M, Issue 1, dated 28.02.08 (*) 2. LHT Engineering Bulletin ref. Doc. No. A320-EB25-1227, Rev. IR, dated 02.06.2007 (*) or later revisions of the above referenced documents | | 28.02.2008 |
| 10014511 | LUFTHANSA TECHNIK AG | EASA.A.S.02586 | | | BOEING 777-200 SERIES | Installation of a stretcher in the passenger compartment for Kuwait Airways | | LHT STC Definition Document - STDD25-06/10M issue 01 rev 01 dated 19 September 2007, or later approved issue. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 06.11.2007 |
| 10014512 | LUFTHANSA TECHNIK AG | EASA.A.S.02587 | | | BOEING 747-400 | Installation of a new stretcher for Kuwait Airlines B747-400 in accordance with Supplemental Type Definition Document STDD25-06 09M Iss. 01, Rev. 01 or later EASA approved revision. | | Supplemental Type Definition Document for B747-400 Stretcher installation KAC STDD25-06_09M Iss. 1, Rev. 1 LHT Engineering Bulletin for the Design, Engineering and Installation of Stretcher for Kuwait Airways B747-EB25-1230 Rev. 1** or later approved | | 09.11.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|----------------|----------------|----------------------|--|---|---|---|---|---------------|
| 10014513 | LUFTHANSA TECHNIK AG | EASA.A.S.02588 | | | AIRBUS A340-300 | Installation of a new Stretcher for Kuwait Airways | | - Supplemental Type Definition Document, Ref. STDD25-06/08M, Issue 1, Rev. 01 dated 19 September 2007; - LHT Engineering Bulletin, Ref. A340-EB25-1228, Rev. 01 dated 18 September 2007; or later approved revisions of the above referenced documents | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 10.10.2007 |
| 10014521 | LUFTHANSA TECHNIK AG | EASA.A.S.02603 | REV 1 | | AIRBUS A300-600 | A300-600 Cabin Reconfiguration from Current Continental Version to Intercontinental Layout | | Compliance Record List CRL25-06/13M, Issue 2 dated 30 May 2006; Supplemental Type Definition Document STDD25-06/13M, Issue 1 dated 6 June 2006 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.06.2006 |
| 10014552 | LUFTHANSA TECHNIK AG | EASA.A.S.02648 | | | AIRBUS A310-300 | A310-300 Siberia Airlines Cabin Reconfiguration from 248 to 255 pax | | Supplemental Type Definition Document STDD25-06/04M, issue 01 dated 22.06.2006; Applicant's Declaration of Compliance dated 22.06.2006 | | 23.06.2006 |
| 10014563 | LUFTHANSA TECHNIK AG | EASA.A.S.02663 | REV. 1 | | A300 SERIES, A310 SERIES | Installation of non-electrical Floor Path Marking System Guideline® | | 1. Supplemental Type Definition Document STDD33-06/02M, issue 01 dated 16.05.20062. Airplane Flight Manual Supplement A-33-06/264-AFM3. Applicant's Declaration of Compliance dated 20.06.2006 or later revisions of the above listed documents approved by EASA | | 14.09.2012 |
| 10014574 | LUFTHANSA TECHNIK AG | EASA.A.S.02678 | | | AIRBUS A319-115 | Interior and systems installation incl. convertible cabin configuration | | Supplement Type Definition document ref. STDD00-06/04M / Issue 2 / dated January 16, 2007. Airplane Flight Manual Supplement ref. A-00-06/444-AFM revision 02, dated January 9, 2007 or later approved revision. | Instruction for Continued Airworthiness, Airworthiness Limitations, Section 4.3, ref. C-AAA000-ICA, issue 2, dated January 12, 2007 or later approved revision | 26.01.2007 |
| 10014593 | LUFTHANSA TECHNIK AG | EASA.A.S.02705 | | | BOEING 737-700 BOEING 737-700 IGW | Upgrade Package of existing VIP Cabin Configuration from private Operation to Commercial Operation | | 1. CP25_06_17M Iss 32. CRL25_06_17M Iss 13. STDD 00_00_02M Iss 04 Rev 06, or later approved revision4. STDD25_06_17M Iss 01 Rev 00, or later approved revision5. A_00_01_311_AFM Supplement Cabin TT&L6. A_00_01_311_AMM Supplement Rev (...) | As indicated in the applicable AFM Supplement- A_00_01_311_AFM Supplement Cabin TT&L for Private Operation- A_25_06_17M_AFM Supplement Cabin TT&L for Commercial Operation. And as indicated in the LBA STC LBA.21E2.TA0307 Compatibility of this de (...) | 18.08.2006 |
| 10014602 | LUFTHANSA TECHNIK AG | EASA.A.S.02725 | | | BOEING 747-100 BOEING 747-200 BOEING 747-300 | Installation of Pneumatic Metric Altimeter | | - Supplemental Type Definition Document No. STDD34-06/01M, Issue 1, Rev. 00, dated Aug 22/06 or later approved revision- Airplane Flight Manual Supplement A-34-06/069-AFM, Revision IR, dated Aug 22/06 or later approved revision | Certification Basis of the Change: CS-25 Amdt. 1 | 25.08.2006 |
| 10014606 | LUFTHANSA TECHNIK AG | EASA.A.S.02730 | | | BOEING MD 11 | Installation of a Rockwell Collins Tailwind 550 Live TV system in Boeing MD-11 as detailed in STDD25-06/19M Supplemental Type Definition Document for Tailwind 550 Live TV installation MD-11 | | 1. A-00-07/510-AFM Airplane Flight Manual Supplement for TW550 Live TV installation, Rev. 01 dated November 06, 20072. R-00-07/535-ICA Instructions for Continued Airworthiness | | 09.11.2007 |
| 10014620 | LUFTHANSA TECHNIK AG | EASA.A.S.02752 | REV. 5 | | B737-700 IGW | Revision of EASA.A.S.02752 Rev.4 per Major Change 0010002741-001: Operation at 8.99 psi maximum cabin differential pressure (Lower Cabin Altitude, LCA) Original STC Title: Design and Implementation of VIP interior and miscellaneous aircraft systems. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI A-01, dated 6th March 2008 plus CRI D-01 (ESF for Width of Aisle). The certificated (...) | - Supplemental Type Definition Doc STDD00-06/10M Issue 03, dated Oct 22, 2012; - AFM Supplement Doc A-00-07/784-AFM, Revision 10, dated Oct 21/11; - Instructions for Continued Airworthiness (ICA) Doc A-00-07/784-ICA Issue 02<->-> dated April 26, 2010; (...) | Once LCA is activated (installation of FAA STC ST01697SE as validated per EASA STC 10042295) on an individual aircraft, the maintenance requirements defined in Doc A-00-07/784-ICA will apply for the life of this aircraft. This limitation remains valid a (...) | 03.12.2012 |
| 10014651 | LUFTHANSA TECHNIK AG | EASA.A.S.02806 | | | AIRBUS A300 | A300F4-605R Multi Mode Receiver (MMR) and Global Positioning System (GPS) installation | | Supplemental Type Definition Document STDD34-06/04M, Issue 1 dated 12 March 2007; Compliance Record List CRL34-06/04M, Issue 1 dated 15 March 2007; Aircraft Flight Manual Supplement A-00-06/460-AFM/ISS, Initial Release dated 01 November 2006; Master Minin (...) | | 22.03.2007 |
| 10014712 | LUFTHANSA TECHNIK AG | EASA.A.S.02885 | | | AIRBUS A300 MODEL B4-603 | A300B4-603 Cabin Reconfiguration from current versions to Intercontinental Layout Cabin Reconfiguration from current versions (total 280 and 232 Passengers) to Intercontinental Layout (total 217 Passengers). | | Compliance Record List CRL25-06/24M, Issue 1 dated 22 January 2007; Supplemental Type Definition Document STDD25-06/24M, Issue 2 dated 22 January 2007 | This STC is approved only for the product configuration as defined in the approved design data referred to in paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.01.2007 |
| 10014713 | LUFTHANSA TECHNIK AG | EASA.A.S.02886 | REV. 1 | | B757-200, B757-300 | Installation of non-electrical Floor Path Marking System Guideline® | | Supplemental Type Definition Document STDD33-06/03M Issue 01 24/11/2006 or later revisions of the above listed documents approved by EASA | Lufthansa Technik AG AFM Supplement A-33-06/472-AFM | 14.09.2012 |
| 10014715 | LUFTHANSA TECHNIK AG | EASA.A.S.02890 | | | BOEING 767-300 | Zonal Drying System Installation | | Supplemental Type Definition Document STDD21-06-02M, Issue 01 rev 00, Feb 01/2008. Instructions for Continued Airworthiness CS21-0200-02 Rev IR, Nov 05/2007 | | 11.02.2008 |
| 10014720 | LUFTHANSA TECHNIK AG | EASA.A.S.02897 | | | A330-300 | Removal of Antenna and Receiver Transmit System and Network Services | | - Supplemental Type Definition Document, Ref. STDD06/15M, Issue 1, Rev. 01 dated 31 May 2007; Instructions for Continued Airworthiness, Ref. ICA-R-00-07/062, Rev. IR dated 10 April 2007 | The AFM Supplement Doc. No. A-00-04/5856 can be removed with the installation of this STC. | 31.05.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|----------------|----------------|----------------------|--|--|---------------------|--|--|---------------|
| 10014721 | LUFTHANSA TECHNIK AG | EASA.A.S.02898 | | | AIRBUS A340-300 | Removal of Antenna and Receiver Transmit System and Network Services | | Supplemental Type Definition Document, Ref. STDD00-06/16M, Issue 01, Rev. 00 dated 1 June 2007. Instructions for Continued Airworthiness, Ref. R-00-07/064, Rev. IR dated 10 April 2007. | The AFM Supplement Doc. No. A-00-04/072 can be removed with the installation of this STC. | 19.06.2007 |
| 10014722 | LUFTHANSA TECHNIK AG | EASA.A.S.02899 | | | AIRBUS A340-600 | Removal of Antenna and Receiver Transmit System and Network Services. | | Supplemental Type Definition Document, Ref. STDD00-06/17M, Issue 01, Rev. 00 dated 1 June 2007. Instructions for Continued Airworthiness, Ref. R-00-07/065, Rev. IR dated 10 April 2007. | The AFM Supplement Doc. No. A-00-04/133 can be removed with the installation of this STC. | 19.06.2007 |
| 10014727 | LUFTHANSA TECHNIK AG | EASA.A.S.02905 | REV. 1 | | ATR 42-200- 300- 320 400- 500 ATR 72-101- 102- 201- 202- 211 ATR 72-212-212A | Installation of non-electrical Floor Path Marking System Guideline® | | -Supplemental Type Definition Document STDD33-06/04M, Issue 01-Certification Program CP33-06/04M, Issue 01 Compliance Record List CRL33-06/04M, Issue 01- Airplane Flight Manual Supplement AFMS A-33-06/497-AFM Initial Revision-Instructions for Continue (...) | ATR42/72 except those aircraft with convertible seats which do not have a fixed lateral position. See Airplane Flight Manual Supplement AFMS A-33-06/497-AFM Initial Revision or later approved revisions. Instructions for Continued Airworthiness ICA33-06 (...) | 14.09.2012 |
| 10014743 | LUFTHANSA TECHNIK AG | EASA.A.S.02926 | | | BOMBARDIER CL-600-2B19 | CL-600-2B19 Cabin Completion - 13 PAX | | Supplemental Type Definition Document STDD00-06/11M, Issue 01, dated September 18, 2007 or later approved revision. Airplane Flight Manual Supplement A-00-07/590-AFM, Issue 01, dated June 13, 2007 or later approved revision. Instructions for Continued (...) | 1. Certification Basis of Change: JAR-25 Change 13, additionally-CRI D-01, Side-Facing Divan-CRI D-04, Door between passenger compartment with integrated blocking functionality-CRI D-05, Isolated compartment-CRI F-01, Executive Power System-JAA TGM Leaf (...) | 19.09.2007 |
| 10014744 | LUFTHANSA TECHNIK AG | EASA.A.S.02927 | | | AIRBUS A300 MODEL B4-600 | Zonal Drying System Installation | | Supplemental Type Definition Document STDD21-06/03M issue 02 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 02.07.2007 |
| 10014801 | LUFTHANSA TECHNIK AG | EASA.A.S.03004 | | | BOEING 737-700 BOEING 737-800 BOEING 737-900 | ATC Mode S # Enhanced Surveillance Upgrade | | - Supplemental Type Design Definition ST DD34-06/C3M Issue 1, Rev 1, dated 26th February 2007, or later EASA approved revision.- AFM Supplement LHT Doc number A-34-07/004-AFM, or later EASA approved revision.- Instructions for Continued Airworthiness Doc number R-34-07/040. | Compatibility of this design change with other previously approved modifications must be determined by the installer. | 02.03.2007 |
| 10014808 | LUFTHANSA TECHNIK AG | EASA.A.S.03012 | REV. 1 | | CL-600-2B19/-2D24/-2D15 | Installation of non-electrical Floor Path Marking System Guideline® | | CRJ-EB33-0320 Revision 1 dated 11 April 2007 AFM A-33-07/012-AFM Revision Initial dated 5 January 2007 or later revisions of the above listed documents approved by EASA | | 14.09.2012 |
| 10014833 | LUFTHANSA TECHNIK AG | EASA.A.S.03048 | | | LEARJET 31A | Navigation # Retrofit of Dual FMC | | Compliance Record List CRL 34-07/01M, Issue 1 (rev. 0) AFM Supplement LBAS310060301, Issue 1 Instruction for Continued Airworthiness LBAS310071201, Issue IR (rev. 0) | None | 02.04.2007 |
| 10014869 | LUFTHANSA TECHNIK AG | EASA.A.S.03094 | | | BOMBARDIER CL-600-2B19 | Cabin Completion - 15 Passengers | | - Lufthansa Technik Supplement Type Definition Document No. STD00-07/08M Issue 1 dated 26 February 2008 or later approved revision- Airplane Flight Manual Supplement No. A-00-07/694-AFM Issue 01 or later approved revision | 1. Installation of "CL-600-2B19-Installation of Structural Provisions for Cabin Completion" approved with EASA STC No. EASA.A.S.03097, is required prior to installation of this STC.2. The Limitation "Installation of Ferry-Kit M/S TC 601R16990" (...) | 11.03.2008 |
| 10014870 | LUFTHANSA TECHNIK AG | EASA.A.S.03095 | | | BOMBARDIER CL-600-2B19 | Cabin Completion | | Lufthansa Technik Supplemental Type Definition Document No. STDD00-07/07M Issue 01 dated 13/02/08 or later approved revision. Airplane Flight Manual Supplement No. A-00-07/693-AFM Issue 01 or later approved revision. | 1. Installation of "CL-600-2B19 - Installation of structural provisions for cabin completion", approved with EASA STC No. EASA.A.S.03097 is required prior to the installation of this STC.2. The limitation "Installation of Ferry-Kit M/S TC 601R (...) | 27.02.2008 |
| 10014871 | LUFTHANSA TECHNIK AG | EASA.A.S.03096 | | | BOMBARDIER CL-600-2B19 | Installation of Structural Provisions for AeroH Antenna system | | LHT Declaration of Compliance No. 1613 dated 22 January 2008 LHT Supplemental Type Definition Document No. STDD-07/06M dated 22 January 2008 | 1. For all flights with "Green Aircraft" and "Structural Provisions for AeroH Antenna System" (2,5 [kg]) installed the installation of Ferry-Kit M/S TC 601R16990 (867 [kg]) is necessary.2. With the installation of Ferry-Kit M/S TC 601R16990(867 [kg]) (...) | 05.02.2008 |
| 10014872 | LUFTHANSA TECHNIK AG | EASA.A.S.03097 | | | BOMBARDIER CL-600-2B19 | Installation of Structural Provisions for Cabin Completion | | LHT Declaration of Compliance No. 1611 dated 29 January 2008 LHT Supplemental Type Definition Document No. STDD00-07/05M, Issue 1 dated 29 January 2008 | 1. For all flights with "Green Aircraft" and "Structural Provisions for cabin completion" (293,7 [kg]) installed the installation of Ferry-Kit M/S TC 601R16990 (867 [kg]) is necessary.2. With the installation of Ferry-Kit M/S TC 601R16990 (867 [kg]) the aircraft will stay in C.G. envelope. | 05.02.2008 |
| 10014873 | LUFTHANSA TECHNIK AG | EASA.A.S.03098 | | | AIRBUS A330-200 | A330 - Installation of Zonal Drying System /The system is installed to get rid of the humidity between the cabin lining and the fuselage | | - Supplemental Type Definition Document, Ref. STDD21-07/05M, Issue 01, dated 25 May 2007- Instructions for Continued Airworthiness, Ref. CS 21-0186-02, Issue 01, dated 17 April 2007 | | 25.05.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|----------------|----------------|----------------------|---------------------|---|---|--|--|---------------|
| 10014893 | LUFTHANSA TECHNIK AG | EASA.A.S.03123 | REV 1 | | AIRBUS A319-100 | PTU and/or Stretcher Installation on A319-100 /Optional replacement of seats by up to 2 Passenger Transport Units (PTU)and/or up to 4 stretchers. | | - Supplemental Type Definition Document, ref. Doc. No. STDD25-07/01M, Issue 1, dated 20.09.07, or later approved revisions- Airplane Flight Manual Supplement, ref. Doc. No. A-00-07/641-AFM, dated 20.09.07, or later approved revisions- Ins (...) | # Detailed visual inspection of the Oxygen Discharge Plug at an interval of 750 FC (C check), as reported in the document Instruction for Continued Airworthiness, ref. Doc. A-00-07/641-ICA Issue 03, dated 19.09.07 | 01.10.2007 |
| 10014897 | LUFTHANSA TECHNIK AG | EASA.A.S.03127 | REV. 1 | | B777-200, B777-300 | Installation of non-electrical Floor Path Marking System Guideline® | | LHT STC Definition Document - STDD33-07/03M issue 02 rev 00dated 23November 2007,or later revisions of the above listed documents approved by EASA | | 14.09.2012 |
| 10014906 | LUFTHANSA TECHNIK AG | EASA.A.S.03142 | REV. 6 | | A318-112 | Structural Floor Panel Reconfiguration | Type Certification Basis in accordance with the TCDS No. A.064 plus- CRI D-01 "Door between passenger compartments, with integrated blockingfunctionality"- CRI D-03 "Isolated Compartments"- CRI D-04 "Firm Handhold"- CRI F-01 "Executive Power System" (...) | - Supplemental Type Definition Document, Doc. No. STD00-07/13M, issued08, dated 28 September 2009, or later approved revisions- Aeroplane Flight Manual Supplement A-00-07/702-AFM revision 6, dated30 September 2008, or later approved revision- Instru (...) | None | 07.10.2009 |
| 10014934 | LUFTHANSA TECHNIK AG | EASA.A.S.03189 | REV 5 | | AIRBUS A318-112 | A318 ELITE # RAK # Structural Floor Panel # SFP - Retrofit | | # Supplemental Type Definition Document, ref. Doc. No. STDD00-07/14M, issue 09, dated 08 April 2009, or later approved revisions# Aeroplane Flight Manual Supplement A-00-07/703-AFM revision 5, dated 30. September 2008, or later approved revisio (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification andany other previously installed modification willintroduce no adverse effect upon the airworthiness of the product.2.Th (...) | 08.04.2009 |
| 10014972 | LUFTHANSA TECHNIK AG | EASA.A.S.03253 | | | AIRBUS A340-212 | WLAN InstallationDesign and implementation of Wireless Local Area Network System | | ¿ Supplemental Type Definition Document, ref. Doc. No. STDD23-07/05M, Issue 1, dated 10.10.07 or later approved revisions¿ Airplane Flight Manual Supplement, ref. Doc. No. A-00-07/705-AFM, dated 26.09.07 or later approved revisions¿ Instructi (...) | | 11.10.2007 |
| 10015011 | LUFTHANSA TECHNIK AG | EASA.A.S.03313 | REV. 1 | | EMB-135/-145 SERIES | Installation of non-electrical Floor Path Marking System Guideline® | | Supplemental Type Definition Document STDD33-07/05M, issue 02.Airplane Flight Manual Supplement A-33-07/709-AFM.Applicant's Declaration of Compliance dated 20.11.2007,or later revisions of the above listed documents approved by EASA | | 14.09.2012 |
| 10015043 | LUFTHANSA TECHNIK AG | EASA.A.S.03359 | | | A300B4-603 / -605R | A300B4-603 / -605R Cabin ReconfigurationCabin Reconfiguration from current version (280 MC) to two possible versions (58 CC / 188 MC or 40 CC / 220 MC). | | Supplement Type Definition Document STDD25-07/10M Issue 01,dated 29 February 2008.Engineering Orders :- EO 156890 Issue 04- EO 156889 Issue 04- EO 156831 Issue 03(or anylater revision approved revision of this document). | The installation of this STC has a prerequisite aircraft configuration.2 This STC is only applicable on aircraft in association with the EASA STC ref: EASA.A.S.02663.3.The Certification Review Item A-01 has to be considered, inaddition to the EASA Certification Basis.4. Prior to inst | 04.03.2008 |
| 10015047 | LUFTHANSA TECHNIK AG | EASA.A.S.03363 | | | BOEING 767-300 | Condor B767 Interior Modification | | Supplemental Type Definition Document, STDD00-07/20M, Issue01, Rev 00, July 01/2008 | # Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. (...) | 29.08.2008 |
| 10015072 | LUFTHANSA TECHNIK AG | EASA.A.S.03402 | | | AIRBUS A318-112 | A318 ELITE - Project BAA; VIP Cabin completion. | | 1. Supplemental Type Definition Document, ref. Doc. No. STDD00-07/21M, issue 03, dated 07. November 2008, or later approved revision.2. Aeroplane Flight Manual Supplement A-00-08/697-AFM issue 1, dated 15. October 2008, or later approved revis (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 07.11.2008 |
| 10015091 | LUFTHANSA TECHNIK AG | EASA.A.S.03428 | | | A3118 | A318 # Ballast Weight Installation | | - Supplemental Type Definition Document, ref. Doc. No. STDD53-08/02M issue 01, dated 11th August 2008 or later approved revisions- Compliance Record List, ref. Doc. No.CRL53-08/02M, issue 01, dated 11th August 2008- Applicant's Declara (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 12.08.2008 |
| 10015107 | LUFTHANSA TECHNIK AG | EASA.A.S.03450 | | | A300 F4-600R | Structure modification for GPS antennae installationThis STC defines the structural modification for the installation of a dual GPS antennae, in place of the existing Gatelink antenna | | Supplement Type Definition Document STDD53-08/03M Issue 01,Revision 1, dated 29 April 2008.Engineering Bulletin : A300-EB53-0292 Issue 03(or any later approved revision of these documents).Instruction for Continued Airworthiness, ref : ICA-53-0292 Issue 1 | | 08.05.2008 |
| 10015120 | LUFTHANSA TECHNIK AG | EASA.A.S.03469 | REV. 1 | | D328 SERIES | Installation of non-electrical Floor Path Marking System Guideline® | | Type Definition Document STDD33-08/01M Issue 01Airplane Flight Manual Supplement A-33-08/034-AFM dated JAN 28/08Instructions for Continued Airworthiness ICA33-08/01M dated FEB 04/08 orlater revision,or later revisions of the above listed documents approved by EASA | | 14.09.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|------------------|----------------|-------------------------|------------------------|---|---|---|--|---------------|
| 10015142 | LUFTHANSA TECHNIK AG | EASA.A.S.03502 | REV 1 | LBA STC LBA.21E2.TA0270 | BOMBARDIER CL-600-2B19 | Replacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via LBA STC LBA.21E2.TA0270 due to the potential for combustion of contaminants in the dryer unit. A ne (...) | | STDD21-02/01M, Issue 04STDD21-07/06M, Issue 01, CL-600-EB21-0192AD 2006-0140-E | See related STC LBA.21E2.TA0270 | 05.08.2008 |
| 10015144 | LUFTHANSA TECHNIK AG | EASA.A.S.03504 | REV 1 | LBA STC LBA.21E2.0307 | BOEING 737-700 | Replacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via LBA STC LBA.21E2.0307 due to the potential for combustion of contaminants in the dryer unit. A new (...) | | STDD00-00/02M, Issue 06STDD21-07/06M, Issue 01, B737-EB21-0193AD 2006-0140-E | See related STC LBA.21E2.0307 | 05.08.2008 |
| 10015145 | LUFTHANSA TECHNIK AG | EASA.A.S.03505 | REV 1 | LBA STC LBA.21E2.TA0636 | BOEING 737-800 | Replacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via LBA STC LBA.21E2.TA0636 due to the potential for combustion of contaminants in the dryer unit. A ne (...) | | STDD00-01/07M, Issue 02STDD21-07/06M, Issue 01, B737-EB21-0195AD 2006-0140-E | See related STC LBA.21E2.TA0636 | 05.08.2008 |
| 10015146 | LUFTHANSA TECHNIK AG | EASA.A.S.03506 | REV 1 | LBA STC TA0733 | BOEING 747-400 | Replacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via LBA STC TA0733 due to the potential for combustion of contaminants in the dryer unit. A new dryer (...) | | STDD21-03/03M, Issue 01STDD21-07/06M, Issue 01, B747-EB21-0196AD 2006-0140-E | See related STC TA0733 | 05.08.2008 |
| 10015147 | LUFTHANSA TECHNIK AG | EASA.A.S.03507 | REV 1 | LBA STC LBA.21E2.TA0609 | AIRBUS A330-200 | Replacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via LBA STC LBA.21E2.TA0609 due to the potential for combustion of contaminants in the dryer unit. A ne (...) | | STDD21-02/03M, Issue 03STDD21-07/06M, Issue 01, A330-EB21-0194AD 2006-0140-E | See related STC LBA.21E2.TA0609 | 05.08.2008 |
| 10015158 | LUFTHANSA TECHNIK AG | EASA.A.S.03526 | | | AIRBUS A318-112 | A318 ELITE - Project Comlux 2; VIP Cabin completion | | Supplemental Type Definition Document, ref. Doc. No. STDD00-08/01M, issue 01, dated 10. December 2008, or later approved revision# Aeroplane Flight Manual Supplement A-00-08/772-AFM Issue 1, dated 13. November 2008, or later approved revision# Instructo (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 12.12.2008 |
| 10015177 | LUFTHANSA TECHNIK AG | EASA.A.S.03554 | | | AIRBUS A318-112 | A318-112 - ELITE - Project HKInstallation of a VIP cabin for a maximum of 19 passengers for commercial and non-commercial operation. The design details are described in General Description S-00-08/731. | | # Supplemental Type Definition Document, ref. Doc. No. STDD00-08/03M, issue 01, dated 27 January 2009, or later approved revision# Aeroplane Flight Manual Supplement A-00-08/835-AFM issue 1, dated 16 January 2009, or later approved revision# (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 28.01.2009 |
| 10015215 | LUFTHANSA TECHNIK AG | EASA.A.S.03630 | | | AIRBUS A318-112 | A318-112 - ELITE - Project Tanker 1Installation of a VIP cabin for a maximum of 16 passengers for non-commercial operation only. The design details are described in General Description C-AAA024-SPA. | | # Supplemental Type Definition Document, ref. Doc. No. STDD00-08/07M, issue 01, dated 18 February 2009, or later approved revision# Aeroplane Flight Manual Supplement A-00-08/873-AFM issue 1, dated 22. January 2009, or later approved revision# (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 20.02.2009 |
| 10027345 | LUFTHANSA TECHNIK AG | P-EASA.A.S.03346 | | | CL-600-2D24 | Introduction of Datalink 2000+ | EASA TCDS No. JMA.023The Certification Basis for the original product remains applicable to this certificate/ approvalThe certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | LHT Declaration of Compliance No. 2035 dated 17 September 2009 LHT Supplemental Type Definition Document No. STDD23-07-07M Issue 2 dated 15 September 2009 LHT Airplane Flight Manual Supplement No. A-23-07/0741-AFM Revision 2 dated 06 August 2009 LHT I (...) | Not Applicable | 23.09.2009 |
| 10027505 | LUFTHANSA TECHNIK AG | P-EASA.A.S.03912 | | | FOKKER F27 MARK 050 | Installation of self-illuminating Floor Path Marking System. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval EASA CS-25, Amendment 5 (elect to comply). The certified noise and/ or emissions lev (...) | Supplemental Type Definition Document STDD33-09/01M, Issue 01, Rev. 01 Engineering Bulletin F50-EB33-0343, Revision 01 Airplane Flight Manual Supplement A-00-09/066-AFM, Revision IR Compliance Record List CRL33-09/01M, Issue 01, Rev. 01* or later revisi (...) | Fokker F50 (F27 Mark 050) aircraft series in standard airline cabin configuration described as typical cabin layout in Engineering Bulletin F50-EB33-0343, Rev. 01, Ch. 4.2. Cabin layout with convertible seats (non fixed lateral position may result in c (...) | 06.10.2009 |
| 10028128 | LUFTHANSA TECHNIK AG | P-EASA.A.S.03435 | | | A330-243 | A330-200 - VIP Cabin Interior Completion | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval Supplemental Type Certification Basis CRI A-01 for P-EASA.A.S.03435 The certified noise (...) | Supplemental Type Definition Document, ref. Doc. No. STDD00-08/02M, issue 01, revision 00, dated 16 November 2009 (>,-<)-> Aeroplane Flight Manual Supplement ref. Doc. A-00-09/009-AFM issue 5, revision IR, dated 12. November 2009 (>,-<)-> Instructions for (...) | as per AFMS A-00-09/009-AFM issue 5 as listed in the associated technical documentation | 03.12.2009 |
| 10028238 | LUFTHANSA TECHNIK AG | P-EASA.A.S.03774 | REV. 1 | | A318-112 | A318-112 - Elite 11 - VIP Cabin Completion | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval EASA Supplemental Type Certification Basis CRI A-5301-006 Issue 1- plus voluntarily by a (...) | Supplemental Type Definition Document, ref. Doc. No. STDD00-09/01M, issue 02, dated Apr. 23/ 2010- Airplane Flight Manual Supplement, ref. Doc. No. A-00-09/053-AFMiss.04 rev.01, dated Nov. 25/ 2009- Instructions for Continued Airworthiness, ref. Do (...) | Not applicable | 26.04.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|------------------|----------------|----------------------|------------------|--|---|---|---|---------------|
| 10028693 | LUFTHANSA TECHNIK AG | P-EASA.A.S.03921 | REV. 1 | | B737-300 | Zonal Dryer System Installation in accordance with 737-EB21-0212 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01/Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane Systems" (...) | Supplemental Type Definition Document STDD21-09/01M Iss.02 Engineering Bulletin 737-EB21-0212 Rev.03 or later revisions of the above listed documents approved by EASA | None | 20.10.2010 |
| 10028830 | LUFTHANSA TECHNIK AG | P-EASA.A.S.03761 | REV. 2 | | A319-133 | A319-133 "GAF" VIP Cabin Completion, Revision 2 to incorporate additional CMR in the ICA and MPD Supplement Closure of post certification action item D-08-01. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate. EASA Supplemental Type Certification Basis CRI A-4301-002 Issue 1 plus voluntarily by applicant: (...) | Supplemental Type Definition Document, ref. Doc. No STDD00-08/17M, issue 03, revision 0, dated 6th August 2010-Airplane Flight Manual Supplement, ref. Doc. No. A-00-09/136-AFM issue 01, rev. 01<->->- dated 15th Feb. 2010-Instructions for Continued Ai (...) | None. | 16.08.2010 |
| 10029134 | LUFTHANSA TECHNIK AG | | REV. 1 | | A340-600 | A340-600 Cabin Retrofit Ethad ETD06EASA STC 10029134 revision due to removal of EWIS limitation | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:- Special Condition(s): CRI E-19 "Mini suites"- the following paragraph(s) at a later amendment: CS-25 Amdt. 5, Appendix H (...) | - Supplemental Type Definition Document, ref. Doc. No STDD00-09/20M, issue 1, dated 24th February 2010- Compliance Record List ref. doc. No. CRL 00-09/20M iss.2, dated 24th April 2013- Instructions for Continued Airworthiness, ref. Doc. No.A-25-10/ (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 13.05.2013 |
| 10029677 | LUFTHANSA TECHNIK AG | | | | A310-304 | Fuselage - Structural provisions and Satcom antenna installation, Satcom equipment rack installation and VHF antenna installation. This STC installs Satcom and VHF antennae, as well as the provision for the future support of a Satcom system installation. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 Issue 1 The certificated noise and/ or emissions levels of the original product (...) | Supplement Type Definition Document, ref : STDD53-09/01M, Issue 1, dated 02 March 2010-Engineering Bulletin, ref: A310-EB53-0308 Issue 3-Engineering Bulletin, ref: A310-EB53-0309 Issue 2-Engineering Bulletin, ref: A310-EB53-0311 Issue 1-Instruction (...) | Limited to: MSN 484 | 12.04.2010 |
| 10030208 | LUFTHANSA TECHNIK AG | P-EASA.A.S.03776 | REV. 2 | | A318-112 | Installation of a VIP cabin for a maximum of 19 passengers. The design details are described in General Description C-AA057-SPA. Deletion of the Airworthiness Limitation Item reference 530000-01-S for A318-112 Elite 14 MSN 4169 | EASA TCDS A.064 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI A5301-008 EASA Supplemental Type Certification Basis plus voluntarily (...) | - Supplemental Type Definition Document, ref. Doc. No. STDD00-09/02M, issue 01, revision 01 dated 02 June 2010- Aeroplane Flight Manual Supplement A-00-09/167-AFM issue 1, revision R, dated 19 April 2010<->->- Instructions for Continued Airworthine (...) | None | 19.07.2011 |
| 10030228 | LUFTHANSA TECHNIK AG | | | | A380-841/ -842 | Installation of Class 2 EFB Laptop Cradle | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate: As per TCDS EASA, A.110 and Certification review Item CRI F-57. The requirements for environmental (...) | Supplemental Type Definition Document, STDD46-09/01M, Issue 1 Airplane Flight Manual Supplement, A-00-10/120-AFM, Issue 1 Instructions for Continued Airworthiness, A-00-10/120-ICA, Issue 01 Applicant's Declaration of Compliance No. 2927, dated 02 June 2 (...) | This STC is only intended for the defined Class 2 EFB with Type A and Type B software applications, in accordance with JAA Temporary Guidance Leaflet TGL-36 and FAA Advisory Circular AC120-76A. Operation of Type C software applications in accordance with (...) | 03.06.2010 |
| 10030374 | LUFTHANSA TECHNIK AG | | | | A340-500 | A340-500 Cabin retrofit ETD03 This STC modifies the cabin layout by replacing First Class seats and all associated seat monuments. | EASA TCDS A.015 Issue 16 Special Conditions:- E-19 (Mini Suite)-D-1304-004 (Flight Crew Rest Compartments) The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | Supplemental Type Definition Document, ref STDD00-10/10M Issue 1 Instruction for Continued Airworthiness, ref A00-10/134-ICA Issue 1 or later revisions of the above listed documents approved by EASA | None | 11.06.2010 |
| 10031201 | LUFTHANSA TECHNIK AG | | | | A340-213 | A340-200 - Installation of provisions for Live TV, Satcom, and W-LAN This STC installs several structural provisions for Live TV, Satcom, and W-LAN systems, relocates the existing upper Anti-Collision Light (ACL) and installs an additional upper ACL. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: Elect to Comply:- CS 25.1729 Amdt 08- CS 25 App H 25.5 Amdt 08 Means of Compliance:- F- (...) | Supplemental Type Definition Document, ref : STDD 00-10/07M Issue 1, dated 22 July 2010 Instruction for Continued Airworthiness, ref : A-00-10_131-ICA Issue 1, dated 06 July 2010 or later revisions of the above listed documents approved by EASA | Limited to MSN 151 | 30.07.2010 |
| 10031674 | LUFTHANSA TECHNIK AG | | | | B747-400 | Trial installation of 2 rows of the BE Aerospace V-seats model BE 15.0PN series 1012121 as replacement for 3 rows of Recaro EC seat model 3410 in Section B of the Lufthansa B747-400 reg D-ABVW, in accordance with LHT document STDD00-10/16M Issue 01 | The Certification Basis for the original product remains applicable to this certificate/ approval | None | This STC is limited to Lufthansa B747-400 aircraft sn 629493, reg D-ABVW. The approval for this trial is valid until November 30, 2010. Compliance with CS 25.853 (heat release/smoke density) has not been shown, in accordance with the certification basis T (...) | 03.09.2010 |
| 10032054 | LUFTHANSA TECHNIK AG | | REV. 1 | | MD-11 | Activation of ADS-B Out capability via 1090MHz Extended Squitter in accordance with Lufthansa Supplement Type Definition Document STDD34-10/01M Iss. 2, dated Sep. 20, 2010 or later approved revision | The Certification Basis for the original product remains applicable to this certificate/ approval | Doc. No. A-00-10/153 Rev IR Airplane Flight Manual Supplement Doc. No. A-00-10/153-ICA IR Instructions for Continued Airworthiness or later revisions of the above listed documents approved by EASA | none | 12.10.2010 |
| 10032628 | LUFTHANSA TECHNIK AG | | | | A319, A320, A321 | New European Cabin Modification | Certification Basis in accordance with EASA Type Certificate Data Sheets A.064. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: 1. CRI H- (...) | - Supplemental Type Definition Document, ref. doc. STDD25-10/21M, Issue 1, dated 17th November 2010- Compliance Record List, ref. doc. CRL25-10/21M, Issue 1, dated 17th November 2010- Applicant's Declaration of Compliance No. 3001, dated 18th November (...) | Not applicable | 18.11.2010 |
| 10032691 | LUFTHANSA TECHNIK AG | | | | A340-600 | A340-600 Cabin Retrofit - ETD05 | Certification Basis in accordance with EASA Type Certificate Data Sheets A.015. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: 1. CRI H- (...) | - Supplemental Type Definition Document, ref. doc. STDD00-10/23M, Issue 1, dated 23rd November 2010- Compliance Record List, ref. doc. CRL00-10/23M, Issue 1, dated 23rd November 2010- Applicant's Declaration of Compliance No. 2961, dated 23rd November (...) | Not applicable | 23.11.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|------------------|----------------|----------------------|--|--|--|--|----------------|---------------|
| 10032887 | LUFTHANSA TECHNIK AG | | | | B737-500 | DLH New European Cabin for 737-500 Aircraft. Modification of the Aircraft seat layout to increase the seat capacity from currently 111 seats to 120 new seats, produced by Recaro Aircraft Seating, Germany. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Supplemental Type Definition Document Ref. STDD00-10/03M issue 01 dated 01 December 2010 - Compliance Record Document Ref. CLRO-10/03M dated 01 December 2010 - Applicant's Declaration of Compliance No 2770 dated 02 December 2010 or later revisions of the above listed documents approved by EASA | Not Applicable | 06.12.2010 |
| 10033112 | LUFTHANSA TECHNIK AG | P-EASA.A.S.03762 | | | A340-300 | A340-300 - "GAF" VIP Cabin Interior Completion | EASA TCDS A.015 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI A-2301-002 EASA Supplemental Type Certification Basis for Lufthansa T (...) | - Supplemental Type Definition Document, ref. Doc. No. STDD00-08/15M, issue 01, revision 01, dated 17 December 2010 -> <->-> Aeroplane Flight Manual Supplement ref. Doc. A-00-09/138-AFM issue 1, revision 03 dated 22 November 2010 -> <->-> Instructions 1 (...) | None | 17.12.2010 |
| 10033691 | LUFTHANSA TECHNIK AG | | REV. 1 | | DHC-8-102, -103, -106 DHC-8-201, -202 DHC-8-301, -311, -314, -315 DHC-8-401, -402 | Installation of Photoluminescent Emergency Floor Path Marking System Guidelines | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 25, Change 9 and FAR 25.1301 (a)(4), Amdt. 123 The requirements for environmental protection (...) | Installation must be in accordance with LHT Engineering Bulletin DHC-8-EB33-0363, Rev 3, dated 03.02.2011. Operation must be in accordance with LHT AFM Supplement A-33-10/222-AFM, IR, dated 03.02.2011. Maintenance must be in accordance with LHT Instruct (...) | Not Applicable | 03.02.2011 |
| 10033738 | LUFTHANSA TECHNIK AG | | REV. 2 | | B747-400 | Activation of ADS-B Out capability via 1090MHz Extended Squitter in accordance with Lufthansa Supplemental Type Definition Document STDD34-10/02M Issue 01, Rev 02, dated Feb. 10, 2011 or later approved revision. | The Certification Basis for the original product remains applicable to this certificate/ approval. | Doc. No. A-00-10/225 Rev IR Airplane Flight Manual Supplement Doc. No. A-00-10/225-ICA Issue 1, Rev. IR Instructions for Continued Airworthiness or later revisions of the above listed documents approved by EASA | None | 17.02.2011 |
| 10034322 | LUFTHANSA TECHNIK AG | | | | A330-343 | Thai Airways A330-300 - Seat exchange and electrical adaptation. This STC installs 263 Y class seats, with associated electrical adaptations. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CI, TCDS EASA.A.004 CRI F-4302-004: IFE and ISPS. The requirements for environmental protection (...) | Supplemental Type Definition Document, ref. STDD00-10/29M Issue 01, dated 17 March 2011 or later revisions of the above listed documents approved by EASA | None | 24.03.2011 |
| 10034796 | LUFTHANSA TECHNIK AG | | | | A380-841, A380-842 | DLH Stretcher Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI D-51 Special Condition Stretcher Installations. The requirements for environmental protection (...) | Definition Document, STDD25-10/23M, Issue 1, Rev 0, 15/04/2011 Declaration of Compliance No. 3153, 18/04/2011 or later revisions of the above listed documents approved by EASA | None | 29.04.2011 |
| 10034892 | LUFTHANSA TECHNIK AG | | | | A330-200 | A330-200 Cabin retrofit ETD01 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: EASA TCDS A.004 Issue 25- Special Condition E-1014 Issue 4 (Inflatable Seat Belt) The re (...) | Supplemental Type Definition Document, ref STDD00-10/26M Issue 1 or later revisions of the above listed documents approved by EASA | None | 10.05.2011 |
| 10034913 | LUFTHANSA TECHNIK AG | | | | 747-400 | Upgrade of first class cabin replacing 16 first class seats by 8 modified first class seats and installation of 8 single beds and half size partitions in accordance with STDD25-10/13M Issue 01, rev 00 <->-> d.d. 11-05-2011 | The Certification Basis for the original product remains applicable to this certificate/ Approval | CP25-10_13M LH FC Update Rev 02 Application Declaration of Compliance ADc 3162 d.d. 11-05-2011 or later revisions of the above listed documents approved by EASA | None | 12.05.2011 |
| 10035638 | LUFTHANSA TECHNIK AG | | | | A318-112 | A318-112 - ELITE 16 MSN 4503 Installation of a VIP Cabin Interior Completion. | Certification Basis in accordance with EASA Type Certificate Data Sheets A.064 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI A- (...) | - Supplemental Type Definition Document, ref. Doc. No STDD00-10/24M, Issue 01 revision 02, dated Jul. 01/2011 - Airplane Flight Manual Supplement, ref. Doc. No. A-00-10/258-AFM issue 01 revision 02, dated May 31/2011 - Instructions for Continued Air (...) | Not applicable | 06.07.2011 |
| 10036121 | LUFTHANSA TECHNIK AG | | | | B747-400 SERIES | Replacement of Global Positioning System Sensor Unit and GPS Antenna. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Lufthansa Supplemental Type Definition Document STDD34-10/05M Issue 1, Rev 0, dated Aug. 11, 2011, Compliance Record List CRI34-10/05M Issue 01 Rev 0 dated 11 August 2011, ICA A-00-11/029, ICA Issue 01 Rev IR dated 25 March 2011 or later revisions of the above listed documents approved by EASA. | None. | 19.08.2011 |
| 10037721 | LUFTHANSA TECHNIK AG | | | | A330-301 / -302 / -303 A330-321 / -322 / -323 A330-341 / -342 / -343 | Thai Airways A330-300 - Seat exchange and electrical adaptation. This STC installs 263 Y class seats, with associated electrical adaptations. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: TCDS EASA.A.004 CS 25 Appendix H25.5 (amdt 8), CRI F-3302-002: IFE and ISPS. The (...) | Supplemental Type Definition Document, ref STDD00-11/17M Issue 01, dated 13 December 2011 or later revisions of the above listed documents approved by EASA. | None. | 16.12.2011 |
| 10038488 | LUFTHANSA TECHNIK AG | | | | A340-311 / -312 / -313 | A340-300 GPSSU Replacement. This STC consists in replacing the current Global Positioning System Sensor Unit (GPSSU) by a new standard able to transmit GPS quality information. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: As per EASA TCDS: A.015 Issue 20 Elect To Comply to CS 25 Amdt 8 for EWIS ICA The require (...) | Supplemental Type Definition Document, ref STDD34-11/01M Issue 1, dated 16 February 2012, or later revisions of the above listed documents approved by EASA. | None. | 21.02.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|----------------|----------------|----------------------|---|--|---|--|---|---------------|
| 10038568 | LUFTHANSA TECHNIK AG | | REV. 1 | | 777-200 | Introduction of Tailwind 550 Antenna | The Certification Basis for the original product remains applicable to this certificate/ approval. | STDD53-05/01M Iss. 02, Rev. 02 dtd. 23.Feb.2012-AFM supplement A-00-11/177-AFM Iss./Rev. 02/00, dated Mar 21/12-Instructions for Continued Airworthiness A-00-12/03-ICA Iss. 02 Rev. 00dtd.31.Jan.2012or later revisions of the above listed documents approved by EASA | This modification is limited to MSN 29953.This modification has a temporary life limit of 500 Flight Cycles. | 11.09.2012 |
| 10039182 | LUFTHANSA TECHNIK AG | | REV. 1 | | B737-700 | Cabin Completion "Tiger" | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate.Special Conditions:CRI D-02 Isolated compartmentsCRI D-06 Compartment Interiors Heat Release and (...) | Compliance Record List CRL00-10/27M Issue 1- Aircraft Flight Manual Supplement A-00-10/281/AFM Issue 1- Instructions for Continued Airworthiness A-00-10/281-ICA Issue for later revisions of the above listed documents approved by EASA | Refer to AFM limitations on the use of Portable Electronic devices | 24.04.2012 |
| 10039304 | LUFTHANSA TECHNIK AG | | | | A340-311/-312/-313 | Navigation - ADS-B Out (Automatic Dependent Surveillance Broadcast)Capability on A340-300This STC consists in installing AOS-B Out capability.This surveillance function is based on the aircraft capability to broadcast, without any ground request, air (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:CS25 Amrdt 10.The requirements for environmental protection and the associated certificate (...) | Supplemental Type Definition Document, ref STDD34-11/02M Issue 1, dated 27 February 2012.Aircraft Flight Manual Supplement, ref. A-00-11/159-AFM Issue 01, dated 07 September 2011.Instruction for Continued Airworthiness, ref. A-00-11/159-ICA Issue 01,d (...) | None. | 24.04.2012 |
| 10039353 | LUFTHANSA TECHNIK AG | | REV. 1 | | A319-111/-112/-113/-114/-115 A319-131/-132/-133 A320-111/-211/-212/-214/-215 A320-216/-231/-232/-233 A321-111/-112/-131/-211 A321-212/-213/-231/-232 | Replacement of Litton MK-1 Global Positioning System Sensor Unit (GPSSU) by CMC Electronics GPSSU CMA-5024 (STC revised for validity extension to A319 and A321 models) | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Supplemental Type Definition Document STDD34-11/21M Issue 2 Rev. 2 dated 07.08.2012- Compliance Record List CRL34-11/21M Issue 2 Rev. 2 dated 07.08.2012- Instructions for Continued Airworthiness A-00-12/01-ICA Issue 1 Rev. 2 dated 02.08.2012- Declara (...) | None | 08.08.2012 |
| 10039798 | LUFTHANSA TECHNIK AG | | REV. 1 | | B747-8 | Installation of class 2 EFB docking station with display, cockpit doorsurveillance cameras an taxi aid cameras at vertical stabilizer and aft fuselage belly. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:CS25 Amendment 5 Appendix H25.5 (EWIS)The requirements for environmental protection and (...) | Aircraft Flight Manual Supplement A-00-12-15-AFM - Issue 01, Instructions for Continued Airworthiness A-00-12-23-ICA - Issue 01,Modification Document STDD46-11/13M Issue 01, Certification program CP46-11/13M issue 01, or later revisions of the above listed documents approved by EASA. | Only for EFB installation. No installation of the belly and taxi aid camera in the frame of this STC. | 17.07.2012 |
| 10039911 | LUFTHANSA TECHNIK AG | EASA.A.S.03922 | | | 777-200 AS PER 777-EB-21-0228 | STDD 21-09/02M: B-777-200 - Zonal Drying System Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.CRI H-1 ICA for EWISThe certificated noise and/ or emissions levels of the original pro (...) | STDD 21-09/02M Iss. 01, Rev 00 dtd. May 25, 2012- Instructions for Continued Airworthiness A-00-12/14-ICA Iss. 02 Rev.IR dtd. May25, 2012or later revisions of the above listed documents approved by EASA | none | 01.06.2012 |
| 10040589 | LUFTHANSA TECHNIK AG | | | | B747-8 | Camera installation in the vertical stabilizer for ground manoeuvring service. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Supplemental Type Definition Document STDD55-11/20M Issue 01 (->-<->-)Design Review C-ABJ275-DR rev. 0or later revisions of the above listed documents approved by EASA | none | 12.07.2012 |
| 10040628 | LUFTHANSA TECHNIK AG | | | | 319-133 | A319-133 ANJ VIP Cabin Completion | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.The EASA Certification Basis for this STC are defined in EASA CRI A-01for LHT Project (...) | Lufthansa Technik's documents:- Supplemental Type Definition Document STDD00-11/16M, Issue 2, Revision 00 dated 13th July 2012 and- Airplane Flight Manual (AFM) Supplement A-00-11/172-AFM Issue 1, Revision 01 dated 30th May 2012, and- Instructions for (...) | - This design change approval is applicable on Aircraft model Airbus A319-133 CJ, aircraft serial number 4428. | 16.07.2012 |
| 10041044 | LUFTHANSA TECHNIK AG | | | | 737-800 | VIP Cabin Interior Completion BBJ2 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.Additional requirements are detailed in CRI A-01 Issue 05, Rev 00 or later revisions of the above listed documents approved by EASA. | Supplemental Type Definition Document STDD00-10/12M, Issue 01, Rev 00- Aircraft Flight Manual Supplement A-00-11/006-AFM, Issue 05, Rev 00- Instructions for Continued Airworthiness A-00-11/006-ICA, Issue 05, Rev 00 or later revisions of the above listed documents approved by EASA. | 1) As per AFM limitations A-00-11/006-AFM, Issue 05, Rev 00 - The aircraft is certified for non-commercial use.- Required Crew: in case passengers are on board one (1) cabin attendants required and has to be seated at door 1, left hand side.- Cabi (...) | 17.08.2012 |
| 10041482 | LUFTHANSA TECHNIK AG | | | | A318-112 | A318-100 - ELITE - Project Comlux3. Green aircraft completion to a VIP cabin configuration. | - Supplement Type Definition Document, ref. Doc. No STDD00-08/16M, issue 01, dated 03. Jul. 2009, or later approved revisions - Airplane Flight Manual Supplement, ref. Doc. No. A-00-08/877-AFM issue 01, dated 23. Jun. 2009, or later approved revisions- (...) | - Supplement Type Definition Document, ref. Doc. No STDD00-08/16M, issue 01, dated 03. Jul. 2009, or later approved revisions - Airplane Flight Manual Supplement, ref. Doc. No. A-00-08/877-AFM issue 01, dated 23. Jun. 2009, or later approved revisions- (...) | | 07.07.2009 |
| 10042417 | LUFTHANSA TECHNIK AG | | | | A320-211 | Integration of the W-IFE (Wireless In-Flight Entertainment) system on A320. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Declaration of Compliance ADC 3491 Rev. IR, dated 28/11/2012;- Compliance Record List CRL23-12/31M Issue 1 Rev. 0, dated 27/11/2012;- Supplemental Type Definition Document STDD23-12/31M Issue 1 Rev. 0, dated 28/11/2012;- Airplane Flight Manual Supp (...) | This approval is limited to MSN 0110. | 29.11.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|----------------------|----------------|----------------------|--|--|--|---|---|---------------|
| 10043402 | LUFTHANSA TECHNIK AG | | | | A319-112 | Installation of NEK cabin modification. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -, Declaration of Compliance ADC 3499 Rev. IR, dated 18/01/2013; -, Compliance Record List CRL25-12/30M Issue 1 Rev. 0 dated 18/01/2013; -, Supplemental Type Definition Document STDD25-12/30M Issue 1 Rev. 0 dated 18/01/2013; -, Instructions for Continuation (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 25.01.2013 |
| 10043864 | LUFTHANSA TECHNIK AG | | | | B747-8 | NG Stretcher installation in aft area of the main deck | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: D-103 "Installation of medical stretcher in cabin" The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Description document STDD25-12/32M Issue 01 Rev. 00 Certification Program CP25-12/32M Issue 01 Rev. 00 Compliance Record List CRL25-12/32M Issue 01 Rev. 00 Engineering Bulletin 747-EB25-1536 Rev 0 or later revisions of the above listed documents approved by EASA | Conditions 1. All equipment (stretcher, oxygen bottles, etc.) must be approved against the appropriate qualification standard, if any. 2. The stretcher shall provide an adequate restraining means for the person, using such devices as a shoulder harness, (...) | 01.03.2013 |
| 10045274 | LUFTHANSA TECHNIK AG | | | | A330-300 | DLH A330 IK Retrofit. Description - Summary : installation of new Y-Class, C-Class and First Class seats and lining. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Certification Plan CP25-12/31M iss 1 at rev 2- Supplemental Type Definition Document STDD25-12/31M issue 1 rev 0 (Master document and definition) - ICA, ref A-00-12/40 ICA, rev 01 - Compliance Record List CRL25-12/31M iss 1 at rev 0 or later revisions of the above listed documents approved by EASA. | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 12.06.2013 |
| 10046218 | LUFTHANSA TECHNIK AG | | | | A319-111 / -112 / -113 / -114 A319-115 / -131 / -132 / -133 A320-111 / -211 / -212 A320-214 / -215 / -216 A320-231 / -232 / -233 A321-111 / -112 / -131 / -211 A321-212 / -213 / -231 / -232 | Installation of HelioJet LED cabin light. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -, Supplemental Type Definition Document STDD33-13/19M Issue 1 Rev 0 dated 16/08/2013; -, Compliance Record List CRL33-13/19M Issue 1 Rev 0 dated 16/08/2013; -, Instructions for Continued Airworthiness A-00-13-41-ICA Issue 1 Rev 0 dated 14/08/2013; -, Applic (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 30.08.2013 |
| 10046576 | LUFTHANSA TECHNIK AG | | | | A340-600 | DLH A340-600 Cabin Interior Reconfiguration. Description - Summary : Cabin Reconfiguration by replacing of all Business- and Economy Class Seats and modification of several components, also the inflight Entertainment System. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Certification Program CP00-12/22M iss 1 at rev 4- Supplemental Type Definition Document STDD00-12/22M issue 1 rev 4- ICA, ref A-00-13/07-ICA, Issue 1 rev 0- Compliance Record List CRL00-12/22M iss 2 at rev 0 or later revisions of the above listed documents approved by EASA. | 1) Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product. (...) | 01.10.2013 |
| 10016808 | LUFTRANSPORT AS | EASA, IM, R.S. 01168 | | | EUROCOPTER FRANCE AS 365N SERI | Mast Measurement System / System for Measurement of ground stations fixed radio antenna for correct frequencies as signal effects | | Master Drawing LDO-2500-1-001 and Design Report LDO-2500-1-002 Luftransport A/S "Flight Manual Supplement SA 365 Series, External Mounted Antenna Model SXH 009A" No. LDO-2500-1-011 | None | 12.06.2006 |
| 10029513 | M&D FLUGZEUGBAU GmbH & CO. KG | P-EASA, A.S. 02904 | | | GLASFLUEGEL 304B | Modification of Glasfluegel 304B, S/N 324: retrofitting with winglets -modified wing tips: winglets. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. | STC "Nachruesten mit Winglets/ Retrofitting with Winglets" of sailplane Glasfluegel 304B, S/N 324 by APDOAP EASA, AP 175, Amended Flight Manual for sailplane Glasfluegel 304B S/N 324, issue March 2010, EASA approved. Amended flutter statement by Prof. Dr. N. Niedbal, issued 15 March 2010. | Limited to Glasfluegel 304B, S/N 324. | 29.03.2010 |
| 10014066 | MACCARTHY AVIATION HOLDINGS Ltd. | EASA, A.S. 01153 | | | BOEING 747-400 | Modification of Cabin to Install Zone B Cabin Class Dividers, Zone B Stowage & Dogbox, Door 2 Port & Starboard Dividers & Upper Deck Dogbox. | | - Modification Leaflet 04ML110. | 1. Compliance with the following Requirements has not been established: 1.1. Viewing of all applicable placards & signage to be in compliance with FAR/JAR 25.791.1.2. Weight, centre of Gravity & Weight Distribution, to be in compliance with JAR 25.1519. (...) | 22.04.2005 |
| 10014966 | MACCARTHY AVIATION HOLDINGS Ltd. | EASA, A.S. 03241 | | | BOEING 757 | Installation of Galley 1B and Forward RH Wardrobe. | | MacCarthy Interiors Modification Leaflet 07ML149. | | 04.12.2007 |
| 10028366 | MACCARTHY AVIATION HOLDINGS Ltd. | EASA, A.S. 03910 | | | A320-232 | Installation of Galley G1 and Galley G5 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01/Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane Systems" (...) | Drawing <-&-> Document Control Report doc. N° 09DC089 Issue B dated 21/12/2009- Drawing <-&-> Document Control Report doc. N° 09DC090 Issue B dated 21/12/2009- Declaration of Design and Performance doc. N° 09DD301 Issue B dated 18/12/2009- Declaratio (...) | The STC holder shall provide by 7th June 2010 Instructions for Continued Airworthiness on Electrical Wiring Interconnection System - EWIS ICA - in accordance with Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS". | 22.12.2009 |
| 10032016 | MACCARTHY AVIATION HOLDINGS Ltd. | | REV. 1 | | A319-133 CJ | Revision of EASA STC 10032016. Removal of the G1, G2 & G5 galleys# loads established on initial issue of EASA STC 10032016 date of issue 01.10.2010 granted to MacCarthy Interiors Ltd. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: Certification basis as per TCDS plus - CRI H-01/Special Condition H-01 "Enhanced Airwor (...) | - Drawing & Document Control Report doc. N° 09DC111 Issue C dated 18/10/2010 (Galley G1) - Drawing & Document Control Report doc. N° 09DC112 Issue C dated 18/10/2010 (Galley G2) - Drawing & Document Control Report doc. N° 09DC113 Issue C dated 18/10/2010. (...) | Approval limited to MSN4114. | 25.10.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|----------------|----------------|----------------------|----------------------------|---|--|---|--|---------------|
| 10044460 | MACCARTHY AVIATION HOLDINGS Ltd. | | | | B747-400 | Installation of Avio economy and business class seats with Lumexis FTTSIFE system. Installation of Lumexis Fiber-To-The-Screen (FTTS) In-Seat AVODin-Fight Entertainment System on AVIO "Centaurus" Economy Class en "Perseus" Business Class seats and in (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Certification Programme: 11CP117, issue 4 (18 March, 2013)-Compliance Checklist: 12CC178, issue 2 (March 2013)-Compliance Checklist: 12CC179, issue 2 (March 2013)-Compliance Checklist: 12CC182, issue A (January 2013)-Instructions for Continued Airwo (...) | STC applicable for B747-400 aircraft MSN 26637 and MSN 26638, to be operated by Transaero. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design (...) | 15.04.2013 |
| 10039505 | MACH TECHNICAL DESIGN S.L. | | | | BELL 412 | Surveillance Camera Installation according to MDL-FAA-0110 Rev. B (The modification consists in the installation of a camera to capture images and the associated equipment to process and send them to a ground station.) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Rotorcraft Flight Manual Supplement FMS-FAA-0110, Issue 1, dated 02/03/2012 or later revisions of the above listed documents approved by EASA-MDL-FAA-0110, Rev B dated 07/03/2012-OT-FAA-0110, Rev 00, dated 19/12/2011-DIP-FAA-0110, Rev 0, dated 01/12/2010 | The installation of this STC is compatible with the installation of the TCCA STC SH96-57 (LBA STC RC.1182). Installation of Helix-Access Steps P/N D205-0633-011 in accordance with Dart Aerospace Ltd. Drawing JN-D205-633, Rev A. Prior to installation of (...) | 08.05.2012 |
| 10046041 | MACH TECHNICAL DESIGN S.L. | | | | A119, AW119MKII | Installation of an External Multipurpose Liquid Injection System with the following main components: an external tank with attached pump fitted to the belly of the helicopter, a control box installed in the cockpit and a remote control anchored to the cyclic stick. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS27 Amdt. 2: 27.1, 27.25, 27.27, 27.45, 27.65, 27.141, 27.143, 27.161, 27.171, 27.173, 27.175, 27.177, 27.251, 27.301, 27. (...) | Rotorcraft Flight Manual Supplement, ref. FMS-FAA-0312 Rev. Original, dtd. 13/Jun/2013 or later revisions of the above listed documents approved by EASA Master Document List, ref. MDL-FAA-0312 Rev. D, dtd. 31/07/2013 Orden Técnica, ref. OT-FAA-0312, Rev. D, dtd. 31/07/2013 | When the External Multipurpose Liquid Injection System is installed, the never-exceed speed Vne is limited to the value which results from subtracting 15KIAS from the never-exceed speed established in the Basic Rotorcraft Flight Manual for each combina (...) | 08.08.2013 |
| 10013933 | MADELEC AERO - F9503 | EASA.A.S.00318 | REV 2 | | FOKKER F.28 MARK 0070 | Installation of photoluminescent Floor Proximity Emergency Escape Path Marking system | | AFM Supplement FMS-9802 Maintenance information provided in MPS-9802 Dispatch deviation procedure provided in MES-9802 | 1. Photoluminescent system charge time: 10 minutes. 2. Photoluminescent Tape life time: 8 years. 3. MSN 11290-11296-11323-113294. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph (...) | 08.03.2006 |
| 10013949 | MADELEC AERO - F9503 | EASA.A.S.00426 | REV 3 | | AIRBUS A319-320-321 | Replacement of the powered floor proximity emergency escape path marking system by a FPEPM photoluminescent system | | Certification Document List (CDL) ref. 9908, Issue of February 20, 2006 Airplane Flight Manual Supplement, ref. FMS-9908, Issue 4, dated February 27, 2006 Master Minimum Equipment List Supplement, ref. MES-9908, Issue 3, dated February 27, 2006 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 31.03.2006 |
| 10015268 | MAP AIRCRAFT PART 21 AS | EASA.A.S.03739 | REV 1 | | BOEING 737-400 | Auxiliary Fuel Tank Removal, B737-400. | | Master Drawing MAP019A101 Iss. D. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 16.02.2009 |
| 10017123 | MAP AIRCRAFT PART 21 AS | EASA.R.S.01210 | | | EUROCOPTER FRANCE AS 332L2 | Installation of Polycon wireless communication system in AS332L2 helicopter | | Polycon AS332L2 Installation Master Drawing - MAP001A101 Issue C Polycon Communication Processing Unit Master Configuration List - 004-RL Issue B Aircraft Flight Manual Supplement - MAP001FMS Issue B Service Bulletin - Polycon SB-03 Issue B | Service Bulletin Polycon SB-03 Issue B shall be implemented before installing this STC | 04.12.2006 |
| 10045479 | MARSHALL AVIATION SERVICES LIMITED | | | | PREMIER 1/1A | To introduce an Aviator Swift Broadband Satcom System with wi-fi. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Hawker Beechcraft Ltd Modification SC0458A, Operation in accordance with Hawker Beechcraft Ltd Flight Manual Supplement No 2 Issue 1 to Manual 390-59001-0003. Maintain in accordance with Hawker Beechcraft Ltd ICA document (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 27.06.2013 |
| 10014154 | MARSHALL OF CAMBRIDGE AEROSPACE | EASA.A.S.01275 | | | SOCATA MS 894A | Marshall Aerospace Modification MA 1665. Radio Equipment Upgrade. | | Marshall Aerospace Modification MA 1665. Marshall Aerospace AFM Supplement AFMS/SOC (1665)/05/0566 | | 26.07.2005 |
| 10031215 | MARSHALL OF CAMBRIDGE AEROSPACE | | | | A340-300 | Structural Installation of 2 IFF Antennae | CS 25 Amendment 6 The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Certificate of Design MA 1903, CD/A340(1903)/09/0625 Issue 2 dated 28 July 2010. Instructions for Continued Airworthiness, SBM23-009 or later revisions of the above listed documents approved by EASA | none | 02.08.2010 |
| 10031227 | MARSHALL OF CAMBRIDGE AEROSPACE | | | | A340-300 | Structural Installation of a V/UHF Antenna | CS-25 Amendment 6 The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Certificate of Design MA 1904, CD/A340(1904)/09/0627 Issue 2 dated 28 July 2010. Instructions for Continued Airworthiness, SBM23-010 or later revisions of the above listed documents approved by EASA | none | 03.08.2010 |
| 10035677 | MARSHALL OF CAMBRIDGE AEROSPACE | | REV. 2 | | A318-112 | Installation of an Auxiliary Fuel Centre Tank system on A318-112 aircraft. Major Change to EASA STC 10035677 Rev 1. This Major Change introduces an alteration to the airworthiness limitation regarding installation on aircraft fitted with a Flammability Red (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: Certification Review Item CRI A-5301-009EASA Supplemental Type Certification Basis for (...) | Certificate of Design doc. No. CD/A318(2036)/13/0410 Issue 1 dated 12/06/2013 Airplane Flight Manual Supplement doc. N°AFM/A318(1778)/10/0943 Rev 3 dated 11/11/2011 Instructions for Continuing Airworthiness - Supplement to Maintenance Manuals doc. No. MT (...) | Restricted to A318-112 aircraft with Airbus modifications 151462 and 151387 embodied. Installation on aircraft fitted with a Flammability Reduction System is only approved when Airbus modifications 152986 and 151311 have been embodied on the aircraft. Pr (...) | 27.06.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|-------------------|----------------|----------------------|---|---|--|--|---|---------------|
| 10035729 | MARSHALL OF CAMBRIDGE AEROSPACE | | REV. 2 | | A340-313 | Iberia Airways Interior Reconfiguration | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS25 Amendment 10, except for CS 25.785 (b)(d) requirements which retain the Certificat (...) | Certificate of design CD/A340(1941)/11/0290 Issue 1 or later revisions of the above listed documents approved by EASA | 1. This STC must be installed in conjunction with EASA STC No.EASA.A.S.00654.2. This STC is limited to the aircraft MSN125, 134, 145, 146, 217, 221, and 329 | 16.07.2012 |
| 10036381 | MARSHALL OF CAMBRIDGE AEROSPACE | | | | B737-700 | 737 Medevac Role Change | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS25 Amendment 6-CR1D-01/10000036 Medical EvacuationThe certificated noise and/or emission (...) | Certificate of Design CD/B737(1797)/09/0630 Issue 1 dated 25/05/2011AFM Supplement AFMS/737/MED(1797)/10/0011 Rev.0 dated Jan 2011 or later revisions of the above listed documents approved by EASA | None | 06.09.2011 |
| 10040450 | MARSHALL OF CAMBRIDGE AEROSPACE | | REV. 1 | | A340-312 | This modification intends to:This modification will installSeat Electronics Boxes (SEB) (PanasonicX2) on to modified existing Avio business class seats and will removeexisting Zone Terminators (ZT) (installed under Marshall ModificationMA1951).Modi (...) | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/or emissions levels of the original product areunchanged and remain applicable to this certificate/approval. | Engineering Report ER/A340(1971)/12/0588, dated 12 Oct 2012. Modification Sheet MA 1971 issue 1, pdf, dated 25 Oct 2012, or later revisions of the above listed documents approved by EASA | none | 25.10.2012 |
| 10015787 | MASS SYSTEMS, | EASA.IM.A.S.01843 | | FAA STC ST01708LA | SD3-30 SD3-60 SD3-SHERPA SD3-60 SHERPA | Installation of MASS Systems# fire extinguisher, P/N M57331-009, as a replacement for the Gravinier fire extinguisher, P/N 57331-009in the above-noted airplane model. This fire extinguisher is in accordance with MASS Systems# Drawing No. M57331-009, (...) | | see above | The approval of this change in type design applies to the above-noted airplane model. This approval should not be extended to otherspecific airplanes of these model series on which other than noted previously approved modifications are incorporated, (...) | 16.04.2007 |
| 10015793 | MASS SYSTEMS, | EASA.IM.A.S.01852 | | FAA STC ST00952LA | 1.AEROSPATIALE ATR 72-101 10.BOEING 757-200 SERIES 11.BOEING 767-200 SERIES 12.BOEING 777-200 13.BOEING DC-10-10 14.BOEING DC-9-81/82 15.LOCKHEED 1011-385-1 2.AIRBUS A300 MODEL B2-1A 3.AIRBUS A310-200 4.AIRBUS A310-300 5.AIRBUS A340-211 6.AIRBUS A340-212 7.BOEING 717-200 8.BOEING 737-200 SERIES 9.BOEING 747-400 SERIES AEROSPATIALE ATR 72-102 AEROSPATIALE ATR 72-201 AEROSPATIALE ATR 72-202 AEROSPATIALE ATR 72-212 AEROSPATIALE ATR 72-212A AIRBUS A300 MODEL B2-1C AIRBUS A300 MODEL B2-203 AIRBUS A300 MODEL B2K-3C AIRBUS A300 MODEL B4-600 AIRBUS A330-301 AIRBUS A340-311 AIRBUS A340-312 BOEING 767-300 SERIES BOEING 777-300 SERIES BOEING DC-9-83 (MD-83) BOEING DC-9-87 (MD-87) BOEING MD-88 | Installation of MASS Systems fire extinguisher cartridges as replacement for Walter Kidde Aerospace fire extinguisher cartridge in fire extinguisher bottles | | FAA Approved Model List dated March 10, 2000, or later approved revisions - MASS System Hermetic Cartridges Kidde Master Drawing List No. 30980, Revision A, dated February 16, 2000, or later approved revision | The approval of this change in type design applies to the airplane models noted on the FAA Approved Model ListA copy ofthe STC must be included in the permanent records of the modified aircraft | 04.10.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------|-------------------------|----------------|----------------------|---|--|---|--|--|---------------|
| 10015794 | MASS SYSTEMS, | EASA.IM.A.S.01853 | | FAA STC ST00961LA | 1.AEROSPATIALE ATR 42-200 2.BELL 204B 3.BOEING DC-8-71 4.BOEING DC-9-14 5.CESSNA 401 6.CESSNA 500 7.EMBRAER EMB-120 8.SAAB SF340A 9.SIKORSKY S-76A AEROSPATIALE ATR 42-300 AEROSPATIALE ATR 42-320 AEROSPATIALE ATR 42-500 AEROSPATIALE ATR 72-101 AEROSPATIALE ATR 72-102 AEROSPATIALE ATR 72-201 AEROSPATIALE ATR 72-202 AEROSPATIALE ATR 72-211 AEROSPATIALE ATR 72-212 AEROSPATIALE ATR 72-212A BELL 206A BELL 205A-1 BELL 205B BELL 212 BELL 412 BELL 412 CF BELL 412 EP BOEING DC-8-61 BOEING DC-8-61F BOEING DC-8-63 BOEING DC-8-63F BOEING DC-8-71F BOEING DC-8-72 | Installation of MASS Systems Fire Extinguisher Cartridge as replacement for Walter Kidde Aerospace fire extinguisher cartridges in fire extinguisher bottles installed in the airplane models noted on the EASA approved model list dtd 02 April 2008 (whi (...) | | These fire extinguisher cartridges are in accordance with the MAAS Systems Hermetic Cartridges Kidde Master Drawing List No. 30990-1 "Slug Thrower Cartridges", Rev N/C, dtd 26 June 2000 or later FAA and EASA approved revisions. | | 08.04.2008 |
| 10015795 | MASS SYSTEMS, | EASA.IM.A.S.01854 | | FAA STC ST00885LA | 1. AIRBUS A300 SEE LIST 10. BOEING 737 SERIES 11. BOEING DC-9 SERIES SEE LIS 12. BOEING MD11 SEE LIST 13. BOEING 747SP SEE LIST 14. BOEING 757 SEE LIST 15. BOEING 767 SEE LIST 16. BOMBARDIER CL-600 SEE LIST 17. EMBRAER MB120 MB145 18. JETSTREAM 4101 18. SAAB 340A 340B 2.A319 A310 SEE LIST 3. A330 SEE LIST 4. A340 SEE LIST 5. BOEING 747 SEE LIST 6. FOKKER F.27 MARK 100 SEE LI 7. F.47 MARK 1000 SEE LIST 8. RAYTHEON HB. 125 SERIES 9. BOEING 707 SEE LIST | Installation of Mass Systems, Inc. fire extinguisher cartridges as replacements for Pacific Scientific fire extinguisher cartridges in fire extinguisher bottles installed in the airplane models noted on the EASA Approved Model List dated 24th January. | | Mass Systems Hermetic Cartridges Master Drawing List No. 30989, Revision B, dated 16th November 1999 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and #Associated Technical Documentation#. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 24.01.2007 |
| 10044728 | MAXIMUM PERFORMANCE INC | | | FAA STC SE7734SW | AEIO-540-D SERIES IO-540-C/-D/-J/-N/-R/-T SERIES O-540-A/B/E/F/G/H/J/L SERIES | Modification of engine for operation with ethanol fuel in accordance with Airmotive Engineering Report No. 9003-8, Revision 2, dated March 12, 1990, and Drawing No. AE077, "Engine Modification Data", Revision 3, dated March 8, 1990, for the Lycoming I (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. | Airmotive Engineering Report No. 9003-8, Revision 2, dated March 12, 1990, and Drawing No. AE077, "Engine Modification Data", Revision 3, dated March 8, 1990, for the Lycoming IO-540 series engines; or Report No. AVS O-540-001, dated September 15, 199 (...) | Approved fuels are as follows: (1) Ethanol, ASTM D4806 (2) Ethanol, Brazil National Department of Fuels, Technical Regulation DNC-01/91, Specification for Anhydrous Fuel Ethanol (#AEAC#) or Hydrous Fuel Ethanol (#AEHC#), (3) Ethanol, EN 15376 <(->)> (4) (...) | 02.05.2013 |
| 10027602 | MAX-VIZ Inc. | P- EASA.IM.A.S.02776 | | FAA STC ST01523LA | F900EX, MF50, MF900 | EASA Validation of FAA STC ST01523LA: Installation of Max-Viz Enhanced Vision System (EVS) 1000 in accordance with Master Drawing List Document number 3038-00025 Revision 5, dated 14th October 2008 or later revisions approved by EASA in accordance with E (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Drawing List Document number 3038-00025 Revision 5, dated 14th October 2008- Aeroplane Flight Manual Supplement Document No. 3038-00026, Revision A dated 16 December 2008 for the basic Mystere Falcon 50 aircraft configuration- Aeroplane Flight Manual Supplement Document No. 3038-00027, Revision A | This design change is also applicable on the Mystere Falcon 50EX designation as defined on EASA TCDS A.062 but limited to the serial number aircraft MSN 286. For this MSN 286 aircraft the applicable EASA Flight Manual Supplement is the initial release o (...) | 13.10.2009 |
| 10017126 | MBH TECHNIC'S | EASA.R.S.01214 | | | SA 365 N S/N 6100 | Installation of EMS cabin configuration, in accordance with MBH Technic's STC file n° 01 MBHT-05, original issue dated 29/03/2005, or later EASA approved revision. | | Installation: MBH Technic's STC file n° 01 MBHT-05, original issue dated 29/03/2005, or later EASA approved revision. Instruction for continued airworthiness MBH Technic's maintenance manual supplement n° 365MBHT01052510, edition 1 dated 06/06, or late (...) | This STC is limited to SA 365 N/s/n 6100 and can be installed only by MBH Technic's technical staff in accordance with MBH Technic's STC file n° 01 MBHT-05, original issue dated 29/03/2005, or later EASA approved revision. | 27.02.2007 |
| 10015807 | MCCAULEY PROPELLER SYSTEMS | EASA.IM.A.S.01875 | | FAA STC SA00798CH | CESSNA FR-182 | Installation of a McCauley Propeller Model B3D36C431/80VSA-1 or B3D36C433/80VSA-1, and spinner assembly. | | In accordance with McCauley Technical Report No 873, no revision, dated June 26, 1997, and McCauley Drawing No D-60202, no revision, dated July 10, 1997, or later approved revisions. | FAA Approved Flight Manual Supplement, Technical Report Number 874, dated August 4, 1997, or later approved revision required. Lycoming engine model O-540-J3C5D is eligible with the McCauley Propeller Model B3D36C433/80VSA-1 (S/N FR18200001 through FR (...) | 08.06.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|---------------------|----------------|----------------------|---|---|--|--|--|---------------|
| 10016128 | MCCAULEY PROPELLER SYSTEMS | EASA.IM.A.S.02322 | | FAA STC SA496CH | BEECH 23 | Installation of McCauley Propeller Model B3D36C429/82NPR-6 | | In accordance with McCauley Technical Report No. 844, no revision, dated January 15, 2006 and McCauley Drawing No. D-60165, no revision, dated April 1, 1996, or later approved revision.FAA approved Flight Manual Supplement, Technical Report No. 845, (...) | 1. Aircraft Specifications are the same as listed on Type Certification Data Sheet A1CE, except as follows:Propeller and Propeller Limits:Propeller: McCauley Constant Speed B3D36C429/82NPA-6Number of blades: 3Blade Angles: At 30° station: L (...) | 22.05.2007 |
| 10029619 | MCCAULEY PROPELLER SYSTEMS | P-EASA.IM.A.S.03127 | | FAA STC SA01511W1 | C400 (LC41-550FG) | Installation of McCauley C447 | CS-23.CThe Certification Basis for the original product remains applicable to this certificate/ approvalThis certificate/ approval involves a change to the certificated noiseand/ or emissions levels | -Installation Instructions No 985 Rev (B) including-Airplane Flight Manual Supplement No 988, Rev (R)-Maintenance Manual Supplement No 989 Rev (A)or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2 (...) | As referred in limitations section of the AFM-S | 06.04.2010 |
| 10043935 | MCCAULEY PROPELLER SYSTEMS | | | | LC42-550FG | Installation of McCauley Model D3A34C444/78MLA-0 | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels. | According to McCauley Installation Instructions Doc. No. 969 Rev -, or later RevisionsTo be operated in accordance with McCauley AFM-S Document No. 970, RevIRTo be maintained i.a.w. McCauley ICA Document No. 975, Rev A.or later revisions of the above (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 06.03.2013 |
| 10015461 | MCFARLANE AVIATION INC | EASA.IM.A.S.01207 | | FAA STC SA01074W1 | CESSNA 150, 152, 170, 172, 175 CESSNA 180, 182, 185, 188, 206 CESSNA 207, 210 | Installation of McFarlane Aviation Flap/Roller Assembly Service Kit MCSK100 under FAA STC SA01074W1 | | McFarlane Aviation Drawings: 1057 Rev Original & 1058 Rev Original FAA Approved April 22, 2002 or later FAA Approved Revision | 1. Cessna Service Bulletin SEB95-3 must be complied with before this STC can be installed.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with (...) | 30.05.2005 |
| 10017129 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01219 | REV. 1 | | AB139/ AW139 | AB139 VIP 28 P/N 6AB1KT003-001 Interior Configuration.This STC supersedes EASA STC EASA.R.S.01219. | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noiseand/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement No 6AB1RFM-1-1 | Approval of this change in type design applies only to those AB139 rotorcraft previously equipped with EASA approved Installation of AgustaKit VIP PROVISION P/N 4G2520F00511. | 09.06.2010 |
| 10017130 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01220 | REV. 2 | | AW139 | VIP 29 Interiors 10 seats P/N 6AB1KT004-001. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certificated noiseand/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | RFM Supplement 6AB1RFM-1-2 Revision N/or later revisions of the above listed documents approved by EASA.NDC 6AB1KT-048 Issue B,Owner's Manual P/N 6AB1WNM-2-2 Reissue 2, Revision E. | None. | 16.07.2012 |
| 10017140 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01242 | REV. 1 | | AB139/ AW139 | VIP 37 Interiors P/N 6AB1KT005-001: Cabin interiors modification.This STC supersedes EASA STC EASA.R.S.01242. | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noiseand/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Publication No. 6AB1RFM-1-3 - Supplement to AB139/AW138 RFM- Publication No. 6AB1WNM-2-3 - Owner's Manual | The change in type design applies only to those AB139 rotorcraft previously equipped with Agusta "Kit VIP Cabin Passenger Provision P/N4G2520F00711". | 09.06.2010 |
| 10017149 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01252 | REV. 2 | | A109S, AW109SP | Cabinet Installation P/N 6A10AD001-001.Extension of the STC10017149 to the AW109SP model that is part of the same TC Number EASA.R.005 | The Certification Basis for the original product remains applicable to this certificate/ approval. | Report 6A10RAD-002 Rev. B "Cabin Cabinet Interiors Configuration"Description"Report 6A10AD-001 Issue B "NDC Cabinet Installation 6A10AD001-001"Report 6A10WNM-2-1 Rev. A "Owners manual" or later revisions of the above listed documents approved by EASA" | None | 01.03.2011 |
| 10017150 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01253 | REV. 1 | | AB139/ AW139 | CD Player Installation P/N 6AB1EL017-001 the change consists in the installation of a CD Player into passenger cabin.This STC supersedes EASA STC EASA.R.S.01253. | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noiseand/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | None | The change in type design applies only to those AB139 Rotorcraft previously equipped with Agusta provision P/N 3G2520A00311 | 09.06.2010 |
| 10017151 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01254 | REV. 1 | | AB139/ AW139 | SAT-COM Installation P/N 6AB1EL018-001The change consists in the installation of a Satellite Telephone System on passenger cabin.This STC supersedes EASA STC EASA.R.S.01254. | The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | None | The change in type design applies only to those AB139 Rotorcraft previously equipped with Agusta SAT-COM provision P/N 3G2310F00411 | 19.06.2006 |
| 10017161 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01269 | REV. 1 | | AB139/ AW139 | Cabinet Utility Installation P/N 6AB1AD004-001.This STC supersedes EASA STC EASA.R.S.01269. | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noiseand/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | NDC n. 6AB1AD-001- Utility Cabinet Installation Owner's Manual, document n. 6AB1WNM-2-6 | Cabinet Utility Installation P/N 6AB1AD004-001 is compatible with AgustaS.p.A. cabin arrangements n. 3G2520F00111 and 3G2520F00211. | 09.06.2010 |
| 10017162 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01270 | REV. 1 | | AB139/ AW139 | 115V AC Power Installation.Electrical power connectors installation in the passenger compartment.This STC supersedes EASA STC EASA.R.S.01270. | The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Technical file per S.E.I.NDC n. 6AB1EL-004/III- Drawing n. 6AB1EL026-001- Owner's Manual n. 6AB1WNM-2-9 | None. | 09.06.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|----------------|----------------|----------------------|---------------|---|---|---|---|---------------|
| 10017163 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01271 | REV. 1 | | AB139/ AW139 | ICS Installation. Intercommunication System in the passenger compartment. This STC supersedes EASA STC EASA.R.S.01271. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | Technical file per S.E.I.NDC n. 6AB1EL-003/III - Drawing n. 6AB1EL020-001-Owner's Manual n. 6AB1WNM-2-10 | None. | 09.06.2010 |
| 10017170 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01278 | REV. 2 | | AB139/ AW139 | AB139 VIP 43 Interiors P/N 6AB1AD006-001 The change consists in customizing passenger cabin interiors (upholstering, installing a decorative bulkhead, entertainment devices, a small cabinet and other minor items). This STC supersedes EASA STC EASA.R.S.01278 Revision 1. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | Report nr. 6AB1RST-008 - Stress Analysis; Report nr. 6AB1RAD-038 - Compliance Check List; Report nr. 6AB1RAD-041 - Interiors Configuration Description; Report nr. 6AB1RAD-042 - Drawing Data Set; Report nr. 6AB1RAD-043 - Cabin Safety Evaluation; (...) | This change in type design applies only to those AB139 rotorcrafts previously equipped with EASA approved installation of Agusta kit VIP PROVISION P/N 4G2520F00811. | 09.06.2010 |
| 10017176 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01294 | REV. 1 | | AB139/ AW139 | VIP 55 Interiors P/N 6AB1AD007-001 The change consists in customizing passenger cabin interiors (personalized seat layout, carpet and liners, devices, adding a small cabinet and others minor items). This STC supersedes EASA STC EASA.R.S.01294. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | Publication nr. 6AB1RFM-1-5 - Supplement RFM | The change in type design applies only to those AB139/AW139 rotorcrafts previously equipped with EASA "Kit VIP Provision P/N 4G2520F01311". | 09.06.2010 |
| 10017198 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01341 | REV. 1 | | AB139/ AW139 | VIP 57 Interiors P/N 6AB1AD008-001. This STC supersedes EASA STC EASA.R.S.01341. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | Publication No. 6AB1RFM-1-6 Supplement to AW139 RFM; Publication No. 6AB1WNM-2-12 Owner's Manual | This STC is only applicable to the AB139 and AW139 helicopters equipped with the Agusta provision for VIP Cabin Interiors P/N 4G2520F01511. | 09.06.2010 |
| 10017255 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01461 | REV. 1 | | AB139/ AW139 | Interiors Configuration Vip 131 P/N 6AB1AD009-001. This STC supersedes EASA STC EASA.R.S.01461. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | RFM Supplement No. 6AB1RFM-1-7 Rev. A dated 16 May 2008 or later approved revision; Owner's Manual No. 6AB1WNM-2-14 Rev. A dated 16 May 2008. | Agusta provision P/N 4G2520F03311 must be installed. | 09.06.2010 |
| 10017261 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01470 | REV. 1 | | AB139/ AW139 | AB/AW139 Seat Armrests Installation P/N 6AB1AD011-001. This STC supersedes EASA STC EASA.R.S.01470. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | Owner's Manual for Armrests Installation P/N 6AB1AD011-001 | This STC can only be installed on the AB/AW139 12-seat configuration P/N 3G2520F00111 that embodies seats P/N 3G2520V00131. | 09.06.2010 |
| 10017267 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01486 | REV. 2 | | AB139/ AW139 | VIP 37 Plus Interiors P/N 6AB1AD012-001 and 9 seats configurations. This STC supersedes EASA STC EASA.R.S.01486 Revision 1. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | NDC n. 6AB1AD-024 for the 7 seats configuration - NDC n. 6AB1AD-035 for the 9 seats configuration - Rotorcraft Flight Manual Supplement n. 6AB1RFM-1-8 Rev. C - Owner's Manual P/N 6AB1WNM-2-16 Rev. C | The VIP 37 Plus Interiors Installation P/N 6AB1AD012-001 can only be installed on AB/AW 139 helicopters equipped with Agusta provision P/N 4G2520F03911. | 09.06.2010 |
| 10017290 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01545 | REV. 1 | | AB139/ AW139 | AW 139 VIP 221 Interiors, P/N 6AB1AD013-001. This STC supersedes EASA STC EASA.R.S.01545. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | NDC 6AB1AD-032 Iss. A Owner's manual 6AB1WNM-2-18 Rev. A RFM Supplement 6AB1RFM-1-10 Rev. A | The VIP 221 interiors can only be installed on AB/AW 139 helicopters equipped with Agusta provision P/N 4G2520F00411. | 09.06.2010 |
| 10030095 | MECAER AVIATION GROUP S.p.A. | | | | AB139 / AW139 | AW139 VIP 305 INTERIOR SP/N 6AB1AD015-001 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Supplement n. 6AB1RFM-1-11 to AW139 Rotorcraft Flight Manual. Rev. A or later revisions of the above listed documents approved by EASA 6AB1 RAD-173: "Description, Compliance evidence" Rev. C 6AB1 RAD-174: "Compliance check list" Rev. B 6AB1 RAD-175: "Ca (...) | As per the approved Rotorcraft Flight Manual Supplement quoted above | 21.05.2010 |
| 10030280 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.00484 | REV. 1 | | AB139/ AW139 | Cabin Configuration VIP-4 P/N 6-A0201K10101-003. This STC supersedes EASA STC EASA.R.S.00484. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ (...) | SFM 6-A02RFM-1-8 | Approval of this change in type design applies only to those AB 139 rotorcraft previously equipped with EASA approved installation of Agusta Kit Vip Cabin interior (Namibia) P/N 3G2520F00411. | 09.06.2010 |
| 10030281 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.00485 | REV. 1 | | AB139/ AW139 | Cabin Configuration VIP-3 P/N 6-A0201K10101-001. This STC supersedes EASA STC EASA.R.S.00485. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ (...) | SFM 6-A02RFM-1-7 | Approval of this change in type design applies only to those AB 139 rotorcraft previously equipped with EASA approved installation of Agusta Kit Vip Cabin interior (Namibia) P/N 3G2520F00411. | 09.06.2010 |
| 10030339 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.00486 | REV. 1 | | AB139/ AW139 | Cabin Configuration VIP-2 P/N 6-A0301K10101-003. This STC supersedes EASA STC EASA.R.S.00486. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | SFM 6-A03RFM-1-6. | Approval of this change in type design applies only to those AB 139 rotorcraft previously equipped with EASA approved installation of Agusta Kit Vip Cabin interior (UAE) P/N 3G2520F00511. | 09.06.2010 |
| 10030340 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.00487 | REV. 1 | | AB139/ AW139 | Cabin Configuration VIP-1 P/N 6-A0301K10101-001. This STC supersedes EASA STC EASA.R.S.00487. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | SFM 6-A03RFM-1-7. | Approval of this change in type design applies only to those AB 139 rotorcraft previously equipped with EASA approved installation of Agusta Kit Vip Cabin interior (UAE) P/N 3G2520F00511. | 09.06.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|-------------------|----------------|----------------------|--|---|--|---|--|---------------|
| 10030342 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01226 | REV. 2 | | AB139/ AW139 | FWD Bulkhead P/N 6AB1AD005-001 Installation. This STC supersedes EASA STC EASA.R.S.01226. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Owner's Manual P/N 6AB1WNM-2-4. | None. | 09.06.2010 |
| 10030371 | MECAER AVIATION GROUP S.p.A. | EASA.R.S.01534 | REV. 1 | | AB139/ AW139 | AW139 "Liners" Silens P/N 6AB1AD010-001. This STC supersedes EASA STC EASA.R.S.01534. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Owner's Manual P/N 6AB1WNM-2-17- Supplement to RFM No. 6AB1RFM-1-9 Rev. A | This STC requires the installation of the Kit Hinged Passengers Doors P/N 4G5212F00111. | 09.06.2010 |
| 10031938 | MECAER AVIATION GROUP S.p.A. | | | | AW139 / AB139 | IFEEL Entertainment System Installation P/N 6AB1AD016-001 | The Certification Basis for the original product remains applicable to this certificate/ approval. | 6AB1RAD-176 Rev. C "IFEEL System Description"- 6AB1WNM-2-20A Rev. A "IFEEL Owner's Manual or later revisions of the above listed documents approved by EASA | none | 23.09.2010 |
| 10038915 | MECAER AVIATION GROUP S.p.A. | | | | MBB-BK117 C-2 | EC145 VIP MB Interiors Drawing No. 6EC1000X001-001 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Helicopter BK117 C-2 RFM Supplement 6EC1RFM-1-1 "EC145 VIP MB Interiors Part Number 6EC1000X001-001" Rev. A or later revisions of the above listed documents approved by EASA. Document 6EC1000N003 "EC145 VIP MB Style configuration Description" Revisio (...) | None | 23.03.2012 |
| 10039438 | MECAER AVIATION GROUP S.p.A. | | REV. 1 | | AW109SP | Installation of Cabinet Five Star P/N 6A10AD002-001 in AW109SP. Modification in accordance with MECAER Notice of Design Change NDC6A10AD-003 Issue A dated 24/04/2012. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Owner's Manual for Cabinet Five Star Installation P/N 6A10AD002-001" P/N 6A10WNM-2-2 First Issue Rev. A dated 04/05/2012 or later approved revision. "Notice of Design Change" Ref. NDC 6A10AD-003 Issue A dated 24/04/2012. "AW109SP Cabinet Five Star Com (...) | This STC approval does not include the installation of the in-flight entertainment equipment inside the Cabinet Five Star. | 16.05.2012 |
| 10042023 | MECAER AVIATION GROUP S.p.A. | | | | AW139 | Ballast Installation P/N 6AB1AD018-001 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | NDC 6AB1AD-146 Issue A; Owner's Manual P/N 6AB1WNM-2-21 Revision A or later approved revision. - RFM Supplement 6AB1RFM-1-12 Revision A; or later revisions of the above listed documents approved by EASA. | This STC is only installable on AW139 Helicopters equipped with Agusta/Westland Active Vibration Control System P/N 4G1830F00111 or 4G1830F00211. | 05.11.2012 |
| 10046602 | MEDITERRANEAN AVIATION COMPANY LTD | | REV. 1 | | PA-31 NAVAJO | Introduction of 120 amp. Alternators on Piper PA-31 - Removal of existing Prestolite 70 Amp. Alternators and install National Air Parts 120 Amp. Alternators model P324RN. The mounting on the engine remains original, since the new alternator is a dir (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | MODE00050_MDL Master Data List Rev. 2 dated 9 September 2013. MODE00050_AAFMS Airplane Flight Manual Supplement. MODE00050_ICA Instructions for Continued Airworthiness dated 26 June 2013, initial issue. MODE00050_WB Weight & Balance Information dated 23 Jul (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 08.10.2013 |
| 10015810 | MEGGITT (ADDISON), Inc. | EASA.IM.A.S.01879 | | FAA STC NO ST09494SC | CESSNA 182S CESSNA 182T CESSNA T182T | Installation of an R134a vapour cycle air-conditioning system | | Install in accordance with Keith Products L.P. Approved Flight Manual Supplement (Certification Report) Ref No CR-182-9, Revision E, dated August 30, 2005 or later approved revision. Operate in accordance with Keith Products L.P. Approved Flight Manual (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 10.04.2006 |
| 10016754 | METRO AVIATION, Inc. | EASA.IM.R.S.01083 | | FAA STC SH3880SW | EUROCOPTER DEUTSCHLAND MBB-BK | Installation of a Vapor Cycle Air Conditioner in accordance with FA STC SH3880SW | | FAA approved Flight Manual Supplement EC145 AC FMS | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 31.05.2005 |
| 10016780 | METRO AVIATION, Inc. | EASA.IM.R.S.01120 | | FAA STC SR01606NY-D | EUROCOPTER DEUTSCHLAND MBB-BK | Installation of Spectrolab SX-5 Starburst searchlight according to FAA STC SR01606NY-D | | Flight Manual Supplement FM00703, Rev IR, FAA approved April 15, 2004. Installation Drawing List IDL00703 Instructions for Continued Airworthiness IC00703 | None | 13.09.2006 |
| 10016823 | METRO AVIATION, Inc. | EASA.IM.R.S.01188 | REV. 1 | FAA STC SR09163RC | EUROCOPTER EC 135 | Installation of mechanically driven vapor cycle air condition system i.a.w. FAA STC SR09163RC as amended by 18-06-2007. Major change to EASA STC EASA.IM.R.S.01188, addition of models and incorporation of technical modifications. | | Flight Manual Supplement 135AC-7, Revision D, dated 4 May 2006 or later approved revision or Flight Manual Supplement 135AC-8, Revision IR, dated 18 June 2007 or later approved revision, Instructions for Continued Airworthiness dated 6 May 1997 or later accepted revision | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 14.04.2008 |
| 10016976 | METRO AVIATION, Inc. | EASA.IM.R.S.01377 | | FAA STC SR09230RC | EUROCOPTER DEUTSCHLAND EC-135 | Installation of landing gear skid extensions | | Flight Manual Supplement FLMS 135-110, Rev. B, dated 15.03.2007 or later approved revision. Installation according to Metro Aviation Master Drawing DL 135-110, Revision IR, dated 23.04.1999 or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 26.11.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|-------------------|----------------|----------------------|--|--|--|--|---|---------------|
| 10035563 | MICHELIN NORTH AMERICA, INC. | | | FAA STC ST03642AT | CL-600-2C10, CL-600-2D15 CL-600-2D24 | Installation of Michelin's tire(s) P/N 029-901-0 on the main gear and/or P/N 036-897-0 on the nose landing gear of Bombardier CL-600-2C10, CL-600-2D15, CL-600-2D24 aircraft wheel assembly. | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certificated noise and/ or emissions levels of the original product areunch (...) | FAA STC No. SA03642AT dated 10 March 2010. Aircraft Braking Systems Corporation (ABCS) Component Maintenance Manual (CMM)/AP-943, Revision 1, dated 20 April 2004, AP-794, Revision 1 dated 1 April 2003, AP-795 dated 1 February 2001, and Michelin North Ame (...) | none | 30.06.2011 |
| 10015379 | MICRO AERODYNAMICS, INC. | EASA.IM.A.S.01058 | | FAA STC SA00763SE | PIPER PA-30 PIPER PA-39 | Installation of Vortex Generators on the Wing leading edge, and vertical fin surfaces. Drawing Package MA2122 Rev. IR dated 25 October 1999, or later approved revision AFM Supplement for PA-30 & PA-39 dated 13 July 2005. Installation Manual MA2123 Rev. (...) | | | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.06.2005 |
| 10015979 | MICRO AERODYNAMICS, INC. | EASA.IM.A.S.02131 | | FAA STC SA00887SE | CESSNA 206 CESSNA 206H CESSNA 207 CESSNA 207A CESSNA P206 CESSNA P206A CESSNA P206B CESSNA P206C CESSNA P206D CESSNA P206E CESSNA T206H CESSNA T207 CESSNA T207A CESSNA TP206A CESSNA TP206B CESSNA TP206C CESSNA TP206D CESSNA TP206E CESSNA TU206A CESSNA TU206B CESSNA TU206C CESSNA TU206D CESSNA TU206E CESSNA TU206F CESSNA U206 CESSNA U206A CESSNA U206B CESSNA U206C CESSNA U206D CESSNA U206E CESSNA U206F CESSNA U206G | Installation of vortex generators on the wings, tail surfaces and vertical fin | | In accordance with Micro Aerodynamics Drawing Package MA2146, Revision A, dated March 11, 2004 and installed in accordance with Installation Manual MA2147, Revision IR, dated January 25, 2001 (for 206 series) or MA2196, Revision IR, dated March 11, 2 (...) | 1. If more than Five vortex generators are damaged or missing, the aircraft is not airworthy. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility w (...) | 20.09.2006 |
| 10016088 | MICRO AERODYNAMICS, INC. | EASA.IM.A.S.02275 | | FAA STC SA00831SE | CESSNA 172A-S CESSNA 172RG CESSNA 175A-C CESSNA P172D | Installation of vortex generators on the wings and tail surfaces in accordance with the provisions of AML SA00831SE. | | Installation must be in accordance with FAA Sealed Drawing List Nr. MA2138 Rev IR5/3/00 and Installation Instructions MA2139, Rev IR dated 5/22/00. | Approval of this change in type design applies to the aircraft listed on AML SA00831SE only. This approval should not be extended to other aircraft on which other previously approved modification are incorporated unless it is determined that the rel (...) | 13.04.2007 |
| 10016354 | MICRO AERODYNAMICS, INC. | EASA.IM.A.S.02675 | | FAA STC SA00941SE | CESSNA 210G,H,J,K,L,M,N, CESSNA P210N CESSNA T210G CESSNA T210J, T210H CESSNA T210K CESSNA T210L CESSNA T210M CESSNA T210N | Installation of vortex generators on the wing and tail surfaces. | | Installation must be in accordance with Drawing Package MA2156 Rev. IR dated 18-April-2001 and Instalation Manual MA2157 Rev. IR dated 03-July-2001, or later Approved Revisions. Maintenance must be in accordance with Micro AeroDynamics, IncInstructio (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 27.02.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------|--------------------|----------------|----------------------|--|---|--|---|--|---------------|
| 10016558 | MICRO AERODYNAMICS, INC. | EASA, IM.A.S.02942 | REV 1 | FAA STC SA00834SE | CESSNA 182C CESSNA 182D CESSNA 182E CESSNA 182F CESSNA 182G CESSNA 182H CESSNA 182J CESSNA 182K CESSNA 182L CESSNA 182M CESSNA 182N CESSNA 182P CESSNA 182Q CESSNA 182R CESSNA 182S CESSNA 182T CESSNA F182-P CESSNA F182-Q CESSNA FR-172-E CESSNA FR-172-F CESSNA FR-172-G CESSNA FR-172-H CESSNA FR-172-J CESSNA FR-172-K CESSNA FR-182 CESSNA R172E/F/G/H/J/K CESSNA R182 CESSNA T182T CESSNA TR182 | Installation of vortex generators on the wings and tail surfaces in accordance with the provisions of AML SA00834SE. | | Approval of this change in type design applies to the aircraft listed on AML SA00834SE only. This approval should not be extended to other aircraft on which other previously approved modifications are incorporated unless it is determined that the rel (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 27.10.2008 |
| 10016576 | MICRO AERODYNAMICS, INC. | EASA, IM.A.S.02974 | REV. 1 | FAA STC SA01266SE | PA-32-301, PA-32-301T PA-32-301FT, PA-32-301XTC PA-32R-301(HF), PA-32R-301T PIPER PA-32R-301(SP) | Installation of vortex generators on the wings and tail surfaces in accordance with the provisions of AML SA01266SE. | CS-23 | Installation of vortex generators on the wings and tail surfaces of the above model Aircraft in accordance with Micro AeroDynamics Drawing Package MA2188, Revision A, dated 26 October 2009 and Installation Manual MA2189, Revision B, dated 26 October 20 (...) | If more than five vortex generators are damaged or missing, the Aircraft is not airworthy. | 10.02.2010 |
| 10016663 | MICRO AERODYNAMICS, INC. | EASA, IM.A.S.03126 | REV 1 | FAA STC SA01646SE | CESSNA 172 CESSNA 175 CESSNA 180 CESSNA 180A CESSNA 180B CESSNA 180C CESSNA 180D CESSNA 180E CESSNA 180F CESSNA 180G CESSNA 180H CESSNA 180J CESSNA 180K CESSNA 182 CESSNA 182A CESSNA 182B CESSNA 185 CESSNA 185A CESSNA 185B CESSNA 185C CESSNA 185D CESSNA 185E CESSNA A185E CESSNA A185F | Installation of vortex generators on the wings and tail surfaces in accordance with the provisions of AML SA01646SE. | | Option 1: Installation of vortex generators on the wings, horizontal and vertical tail surfaces of the above model Aircraft in accordance with Micro AeroDynamics Drawing Package MA2208, Revision IR, dated 17 January 2006 and Installation Manual MA22 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The i (...) | 17.03.2009 |
| 10031549 | MICRO AERODYNAMICS, INC. | | | FAA STC SA00834SE | 1. 150F, 150G, 150H, 150J 1. 150K, 150L, 150M, A150K 1. A150L, A150M, 152, A152 2. F150G, F150H, F150J 2. F150K, F150L, F150M, 2. FA150K, FA150L, F152, FA152 | Installation of vortex generators on the wings and vertical tail surfaces on Cessna 150/152 (see applicable Model list). | CS-23.C | Installation of vortex generators on the wings and vertical tail surfaces in accordance with Drawing Package MA2081, Revision IR, dated 15th January 2002, and Installation Manual MA2082, Revision A, dated 17th April 2002 or later approved revision. Main (...) | Approval of this change in type design applies to the aircraft listed on AML SA01065SE only. This approval should not be extended to other aircraft on which other previously approved modifications are incorporated unless it is determined that the relatio (...) | 25.08.2010 |
| 10034885 | MICRO AERODYNAMICS, INC. | | | FAA STC SA02098SE | 65-90, 65-A90, 65-A90-1 65-A90-2, 65-A90-3, 65-A90-4 B90, C90, C90A, C90GT C90GTI, E90, H90 | Installation of Vortex Generators on the wings and tail surfaces, and strakes on the engine nacelles. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Micro AeroDynamics Drawing Package MA2236, Revision IR, dated October 06, 2010, and Installation Manual MA2237, Revision IR, dated December 01, 2010. Instructions for Continued Airworthiness in accordance with MicroAeroD (...) | If more than five vortex generators, or either nacelle stake are damaged or missing, the aircraft is no longer airworthy. | 10.05.2011 |
| 10038573 | MICRO AERODYNAMICS, INC. | | | FAA STC/SA01681SE | 1. PA-34-200, PA-34-200T, 1. PA-34-220T (SENECA III) 1. PA-34-220T (SENECA IV) 2. PA-34-220T (SENECA V) 3. PZL M20 03 "MEWA" | Installation of vortex generators on the wings, horizontal and vertical tail surfaces (all models), and an increase in maximum zero fuel weight (PA-34-200T and M20 03 only). | Design: Non-significant change. The Certification Basis of the original product remains applicable to this change. Environmental Protection: The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with FAA approved Micro AeroDynamics, Inc. Drawing Package MA2212, Revision A, dated 14-Aug-2006 and FAA approved Installation Manual MA2213, Revision A, dated 06-Oct-2006; Micro AeroDynamics POH/AFM Supplement No. MA2215, Re (...) | If more than five vortex generators are damaged or missing, the aircraft is no longer airworthy. | 27.02.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|-------------------|----------------|------------------------|--|--|---|---|--|---------------|
| 10040949 | MICRO AERODYNAMICS, INC. | | REV. 2 | FAA STC SA01161SE | PA-32-260, PA-32-300 PA-32S-300, PA-32R-300 | Installation of Vortex Generators on the wing and tail surface according to drawing package MA2174, revision IR dated 06-Sep-2002 and installation manual MA2175 revision A dated 27-Aug-2003 or later FAA Approved revision to either of these documents. | The Certification Basis of the original product remains applicable to this change. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA approved drawing package MA2174, revision IR dated 06-Sep-2002. FAA Approved installation manual MA2175 revision A dated 27-Aug-2003. Instruction for continued airworthiness MA0666, or later revisions of the above listed documents approved by EASA (...) | If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. If more than five vortex generators are damaged or missing, the A/C is not airworthy. | 11.09.2012 |
| 10041846 | MICRO AERODYNAMICS, INC. | | | FAA STC SA02226SE | DA 42, DA 42NG, DA 42M-NG | Installation of vortex generators on the wings and tail surfaces. FAA STC SA02226SE | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | AFM Supplement MA 2243, dated October 01, 2012 or later revisions of the above listed documents approved by EASA Drawing Package MA2240, Revision IR, September 30, 2011 Installation Manual MA2241, Revision IR, November 21, 2011 or later revisions of the a (...) | None | 19.10.2012 |
| 10043329 | MICRO AERODYNAMICS, INC. | EASA.IM.A.S.03233 | | FAA STC NO SA01555SE | 19A, B19, M19A, 23, A23, A23A A23-19, A23-24, B23, C23 A24, A24R, B24R, C24R | Installation of Vortex Generators on the wings and tail surfaces | | FAA approved Drawing Package, Doc. No. MA2204, Rev. IR, dated 07-Apr-2005, or later approved revision FAA approved Installation Manual, Doc. No. MA2205, Rev. IR, dated 04-May-2005, or later approved revision Instruction for Continued Airworthiness, stated on VG Operating Placard, MA8001 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The (...) | 02.07.2009 |
| 10015935 | MID-CANADA MOD CENTER | EASA.IM.A.S.02075 | | SA03-92 | | Cockpit voice recorder replacement with Universal CVR 30 or CVR 120 | | Avionics Design Services Ltd. Master Drawing List ref. MDL03074 revision B, dated August 6, 2004 | -- | 14.12.2006 |
| 10036039 | MID-CANADA MOD CENTER | | | | CESSNA 550, S550 | Installation of Rockwell Collins Proline 21 Integrated Display System with new VHF communication transceivers, Navigation Receivers with integral ADF, Distance Measuring Equipment Transceivers and Mode S Transponders with Elementary Surveillance capabilities (...) | EASA TCDS IM.A.207 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRJ F-02: Airworthiness Criteria for RNAV Equipment supporting LPV Op (...) | Avionics Design Services Ltd. Master Drawing List MDL09001 Issue 1, Revision B from 29 July 2011 Mid Canada Mod Center Flight Manual Supplement FMS09001-1 Revision A from 29 July 2011 Avionics Design Services Ltd. Instructions for Continued Airworthiness (...) | None | 08.08.2011 |
| 10038891 | MID-CANADA MOD CENTER | | | TCCA STC SA10-16 | CL-600-2B19 | CMC Electronics CMA-1100, Class II Electronic Flight Bag (EFB) System Installation | EASA.IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged (...) | Transport Canada STC No. SA10-16, Issue 1, dated 03 February 2010. Avionics Design Services Ltd Master Drawing List MDL09134, Revision N/C, dated 20 January 2010, or later approved revision Avionics Design Services Ltd Flight Manual Supplement FMS0913 (...) | None | 21.03.2012 |
| 10041743 | MID-CANADA MOD CENTER | | | TCCA STC O-LSA09-348/D | CL-600-2B19 | TrueNorth Satphone Installation | EASA.IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged (...) | Transport Canada STC No. O-LSA09-348/D Issue 2 dated 13 June 2012 Avionics Design Services Ltd. Master Drawing List MDL09014, Revision C, dated 13 June 2012. Mid Canada Mod Center Flight Manual Supplement FMS09014-1, Revision N/C, dated 12 June 2012 or (...) | Restricted to ASN 8092 EASA STC Nos. 10038463 and 10028338 Rev. 2 (Transport Canada STC Nos. O-LSA09-344/D and C-LSA09-326/D) are pre-requisites to this approval. | 12.10.2012 |
| 10046006 | MID-CANADA MOD CENTER | | | TCCA STC SA11-6 | MF50 | TrueNorth Avionics Express WiFi system installation | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Avionics Design Services Ltd Master Drawing List MDL10094 Rev C, dated March 30, 2011 - Mid Canada Mod Center Flight Manual Supplement FMS10094-1 Rev B (> <) - dated March 30, 2011 - Mid Canada Mod Center Instructions for Continued Airworthiness ICA10094 (...) | None | 05.08.2013 |
| 10015681 | MIDCOAST AVIATION INC | EASA.IM.A.S.01663 | | FAA STC ST00807W1 | BAE 125 SERIES 1000A HAWKER 1000 | Installation of Allied Signal Mk-V Enhanced Ground Proximity Warning System (EGPWS) | | In accordance with Midcoast Certification Drawing List No 26-327G001, Rev B, dated July 26, 1999, or later approved revisions. FAA Approved Midcoast Aircraft Flight Manual Supplement Document No 26-327M001, Rev "Original", dated July 29, 1999, or later approved revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 03.03.2006 |
| 10015855 | MIDCOAST AVIATION INC | EASA.IM.A.S.01958 | | FAA STC ST01258WI-D | CL-600-1A11, CL-600-2A12 CL-600-2B16 | Installation of an Audio Intercom Inc. Articulated Mounting Arm, KGS LT-71(J) Power Suppl & Data Connectivity for an Electronic Flight Bag | | Midcoast Aviation Inc Master Data List 34-343G001 Revision A dated July 14, 2004 or later approved revision | The Audio Intercom Interarticulated Arm Attachment is limited to the weight & CG Envelope Contained in Midcoast Aviation Inc. Weight & Balance Update 34-010E092 Revision A dated July 13, 2004 or later approved revision | 17.08.2006 |
| 10015913 | MILLENNIUM CONCEPTS, INC. | EASA.IM.A.S.02042 | REV. 1 | FAA STC SA01160W1 | CESSNA 208 CESSNA 208B | Installation of Executive Interior in Cessna Models 208, 208B Caravan | | In accordance with Millennium Concepts Inc Master Drawing list (MDL) 2015-MDL022, Rev Y, dated May 15 2007, or later approved revisions. | | 08.02.2008 |
| 10041159 | MILLENNIUM CONCEPTS, INC. | | REV. 1 | FAA STC SA01671W1 - D | 1- G36 - BONANZA 2- G58 - BARON | Installation of interior configuration per Master Drawing List 2033-MDL040, rev. P, dated 8 February 2012. | Design: Non-significant change. The Certification Basis of the original product and the following additional airworthiness requirements are applicable to this change. As defined in the STC certification basis approved by FAA. Environmental Protection: The (...) | FAA approved Master Drawing List 2033-MDL040, rev. P, dated 8 February 2012. FAA Approved AFM Supplement 2033-AFMS-S1 dated 8 February 2012 (For G36 Bonanza). FAA Approved AFM Supplement 2033-AFMS-S2 dated 8 February 2012 (For G58 Baron). Instruction for c (...) | If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. | 27.08.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|--------------------|----------------|----------------------|--|--|---|---|--|---------------|
| 10016595 | MITSUBISHI HEAVY INDUSTRIES AMERICA | EASA.I.M.A.S.03006 | | FAA STC SA1751SW | | Flight Crew Restraint System. | | FAA STC SA1751SW | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The i (...) | 22.09.2008 |
| 10038450 | MITSUBISHI HEAVY INDUSTRIES AMERICA | | | FAA STC SA01594SE | 1- MU-2B-20, -25, -26 2- MU-2B-30, -35, -36 3- MU-2B-20, -25, -26 3- MU-2B-30, -35, -36 4- MU-2B-25, -26, -26A 4- MU-2B-35, -36, -36A 4- MU-2B-40, -60 | Fabrication and Installation of Stretched Acrylic Cabin Windows in accordance with FAA STC SA01594SE. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List MU2-11000, Revision C, dated December 28, 2005. Installation and Maintenance Instructions Drwg. MU2-11001, Rev. IR, dated December 28, 2005; or later revisions of the above listed documents approved by EASA in accordance with EASA ED (...) | None. | 17.02.2012 |
| 10014115 | MONARCH AIRCRAFT ENGINEERING LTD | EASA.A.S.01221 | | | BOEING 737-522 | Installation of Honeywell EGPWS | | -Monarch Aircraft Engineering Modification MON/737/34/3013 Issue A dated March 3 2005 or later approved revision-Monarch Aircraft Engineering Flight Manual Supplement number 1 Issue 1 reference MON-34-49-2785 or later approved revision | | 06.04.2005 |
| 10014116 | MONARCH AIRCRAFT ENGINEERING LTD | EASA.A.S.01222 | | | BOEING 737-522 | Introduction of 131 Passenger Configuration | | -Monarch Aircraft Engineering Modification MON/737/25/3016 | | 01.04.2005 |
| 10014155 | MONARCH AIRCRAFT ENGINEERING LTD | EASA.A.S.01277 | | | BOEING 767-300 | Monarch Aircraft Engineering Modification MON/767/258/2995 Installation of a 38 Premium Class Cabin | | Monarch Aircraft Engineering Modification MON/767/25/2995 | Subject AFM supplement is associated to Dassault Aviation Mod M1632: "Installation d'un hublot inférieur avec volet de protection", applicable to F20 GF SN 397 only | 29.06.2005 |
| 10044060 | MONARCH AIRCRAFT ENGINEERING LTD | | | | A319-115, A320-212/-214 A321-231 | Installation of the Wireless IFE System | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amendment 12; CS 25.1703CS 25.1707CS 25.1709CS 25.1711CS 25.1713CS 25.1715CS 25.1717CS 25.1719CS 25.1721CS 25.1729T (...) | Monarch Design Services Modification Approval Document MON/320/23/MAD/3383 Issue B/Monarch Design Services Flight Manual Supplement MON-23-36-3396 Not Issue A or later revisions of the above listed documents approved by EASA | None | 14.03.2013 |
| 10013959 | MOTORFLUG BADEN-BADEN GmbH | EASA.A.S.00610 | | | SOCATA TB 20 | Installation of Transponder Garmin GTX 330 - Mode S | | -AFM Supplement TB 20-01-0601 AFM Supplement TB 20-01-0602-Instructions for Continued Airworthiness TB 20-01-0701 | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 01.03.2005 |
| 10028137 | MOTORFLUG BADEN-BADEN GmbH | P-EASA.A.S.03954 | REV. 2 | | P.180 AVANTI II | Additional installation of Collins BCU-4000 and CMU-4000 to flight inspection system. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23 Amdt. 2 There are requirements for environmental protection and the associated certificate (...) | -Aircraft Flight Manual Supplement 281F93072-0601, Rev. 4.25.11.2011 or later revisions of the above listed documents approved by EASA - Instructions for Continued Airworthiness 281F93072-0701, Rev. 3.20.09.2011- Installation Manual 29131F108371-1001, Rev. 0, 15.09.2011 | None. | 01.10.2012 |
| 10028394 | MOTORFLUG BADEN-BADEN GmbH | | | | PIAGGIO P.180 AVANTI II | Installation of TCAS top & bottom antennas. Installation of TCAS processor in nose section. Interface to existing transponder and avionics system. | CS 23 | Installation in accordance with Master Document List (MDL) 281F93887-0109 Rev. 1, and Installation Manual 281F93887-1001 Rev. 1 <(->-) Operation in accordance with AC Flight Manual Suppl. (AFMS) 281F93887-0601 Rev. 0 <(->-) Instructions for Continued Air (...) | None | 04.01.2010 |
| 10033883 | MOTORFLUG BADEN-BADEN GmbH | | | | 407 | Installation of Antenna Mount on Tailboom | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 28107F103283-1001 revision 0 dated 25 May 2010 - Installation Manual 28107F103283-0701 revision 0 dated 25 May 2010 - Instructions for Continued Airworthiness 28107F103283-0101 revision 0 dated 25 May 2010 - Compliance Program | none | 16.02.2011 |
| 10038604 | MOTORFLUG BADEN-BADEN GmbH | | REV. 1 | | 228-212 | Installation of Mission Equipment for Maritime Surveillance. | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certified noise and/ or emission levels. | 28145F103137-0601, Rev. 2., AFM Supplement 28145F 103137-0701, Rev. 2., Instruction for Continued Airworthiness 28145F103137-0702, Rev. 1., TLMCM Supplement 28145F103137-1001, Rev. 0., Installation Manual or later revisions of the above listed documents approved by EASA. | None. | 11.10.2011 |
| 10041143 | MOTORFLUG BADEN-BADEN GmbH | | | | 204B | Exchange of engine torque and temperature indicator. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 29 change 1 dated 30 November 2007; 29.1, 29.29, 29.305, 29.361, 29.561, 29.601, 29.6 (...) | 127472-01-0601 revision 0 dated 13 August 2012, Flight Manual Supplement Applicable revisions or later revisions of the above listed documents approved by EASA 127472-01-1001 revision 0 dated 13 August 2012 - Installation Manual; 127472-01-0109 revision (...) | none | 23.08.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|----------------------|-------------------------|---|---|---|---|---------------|
| 10043664 | MOTORFLUG BADEN-BADEN GmbH | | REV. 1 | | A109S | Installation of Max-Viz EVS-1500 infrared camera. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The modification is compliant with the requirements of JAR 27 Amendment 3 paragraphs: 25, 29, 65, 67, 141, 143, 171, 173, 1 (...) | RFM Supplement doc. 25105F101554-0601 Revision 0, dated 15.01.2013; or later revisions of the above listed documents approved by EASA. Master Document List doc: 25105F101554-0109 Revision 3 dated 15.01.2013; - Instructions for Continued Airworthiness (...) | None. | 25.02.2013 |
| 10045063 | MOTORFLUG BADEN-BADEN GmbH | | | | A109S | A109S installation of EGPWS MK XXII | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-27 Amdt. 2, paragraphs 25, 29<->-> 301, 303<->-> 305, 307, 561, 60 (...) | RFMS for Installation of Honeywell MK XXII EGPWS, Document No.28107F94078-0601, revision 0, dated 27.02.2013 or later revisions of the above listed documents approved by EASA. Master Document List for Installation of Honeywell MK XXII EGPWS, Document N (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 28.05.2013 |
| 10045303 | MOTORFLUG BADEN-BADEN GmbH | | | | MBB-BK 117 C-2 | Installation of HATS Tracking System, used for flight tracking of the Rotorcraft. The HATS consists of the Tracking Device (HATC), one external installed GSM antenna and one external installed satcom antenna. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | RFMS 25102F130209-0601 Rev 2 dated 11/06/2013 or later revisions of the above listed documents approved by EASA. CP 25102F130209-0101 dated 13/03/2013 MDL 25102F130209-0109 Rev 2 dated 13/06/2013 | During take-off and landing procedure the HATS system audio channel has to be switched off on pilots and co-pilots audio control unit in both reception and transmission control. HATC's maintenance interface shall be used only on the ground for maintenance p (...) | 14.06.2013 |
| 10016537 | MPM INC. | EASA.IM.A.S.02914 | | FAA STC SA01033WI | | Replacement of original main landing gear doors with one piece monolithic machined gear doors. | | Installation in accordance with FAA approved, MPM Inc, Installation Drawing ref. LG-DR002 revision "none" or later approved revisions | 1. The attached AML is applicable only on aircraft serial numbers called out in MPM Inc Drawing No AML-LG-DR002-1, -2, -3 and -4 for installation and applicability. 2. Prior to installation of this modification the installer must determine that (...) | 05.06.2008 |
| 10016538 | MPM INC. | EASA.IM.A.S.02915 | | FAA STC SA00902WI | | Replacement of original main landing gear doors with monolithic machined gear doors. | | Installation in accordance with FAA approved, MPM Inc, Installation Drawing ref. LG-DR0021 and Installation Instructions MPMI-103 on aircraft listed in the attached Approved Model List (AML) in MPM Inc, Drawing Number AML-LG-DR0001, Rev A, dated November 27, 2000, or later approved revision. | 1. The attached AML is applicable only on aircraft serial numbers called out in MPM Inc Drawing No. AML-LG-DR0011. The attached AML is applicable only on aircraft serial numbers called out in MPM Inc Drawing No. AML-LG-DR001.2. Prior to install (...) | 10.06.2008 |
| 10042986 | MTI POLYFAB Inc. | EASA.IM.A.S.01963 | | TCCA STC SA03-19 | CL-600-1A11/-ZA12/-ZB16 | Installation of Insulation Blankets | | - Transport Canada STC SA03-19 Issue 2 dated February 19, 2003<->-> Magnifoam Technology Inc. Modification Summary MS16050201, MS 16050202<->-> MS 16050203, MS16050204, MS 160500501 Revision NC or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.09.2006 |
| 10042987 | MTI POLYFAB Inc. | EASA.IM.A.S.01964 | REV. 1 | TCCA STC SA05-68 | BD700-1A10, BD700-1A11 | Thermal Acoustic Insulation Installation | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA05-68 Issue 3 dated 10 May 2007/MTI-Polyfab Inc. Modification Summaries as listed below or later approved revisions: Model, ..., Modification Summary No. ..., Revision No. BD-700-1A10, ..., MS01010607, ..., NC, BD-700-1a11, ..., MS0 (...) | None | 01.10.2013 |
| 10014045 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.01126 | REV. 1 | | REVO LAKE 250 | Installation of propeller- MTV-9-B/LD193-109 or- MTV-9-B-C-R(M)/CRLD193-109 | | - Anhang zum Flugbuch Nr. E-1198- Airplane Flight Manual Supplement No. E-1199- Umrüst- und Wartungsanweisung E-1194- Instructions for Continued Airworthiness and Installation Instructions E-1197- Anhang Flughandbuch Nr. E-1200- Airplane Flight Man (...) | | 02.12.2005 |
| 10014046 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.01128 | REV. 1 | | CESSNA 208, 208B | 1. Installation of the MTV-16-1-E-C-F-R(P)/CFR250-55 full feathering and reversible 4-blade constant speed propeller. 2. Extension of Installation 1 on MTV-16-1-E-C-F-R(P)/CFR250-55 propellers equipped with fluid device system. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. FAR 23 including amendment 34 (not required by 21A.101, elected to comply by the applica (...) | Installation 1: - Instructions for Continued Airworthiness and Installation Instructions No E-1169- Airplane Flight Manual Supplement No E-1169- Airplane Flight Manual Supplement No E-1168 Installation 2: - Instructions for Continued Airworthiness and Installation Instructions No E-1169, Revision 2-A (...) | None | 12.04.2011 |
| 10014100 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.01202 | | | SOCATA TB 21 | Installation of MTV-9-B/198-52 hydraulically controlled 3-blade constant speed propeller | | - Instructions for Continued Airworthiness and Installation Instructions No E-1189- Airplane Flight Manual Supplement No E-1187- German "Anhang zum Flughandbuch" No E-1186 | | 01.08.2005 |
| 10014107 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.01212 | | | MOONEY M20M | Installation of propeller MTV-14-B/185-59b | | Anhang zum Flughandbuch Nr. E-1221 Airplane Flight Manual Supplement No. E-1222 Umrüst- und Wartungsanweisung E-1223 Instructions for Continued Airworthiness and Installation Instructions E-1224 | | 02.11.2005 |
| 10014194 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.01336 | | | PITTS S-2C S-2B | Installation of propeller MTV-9-B-C/C203-46 | | Pitts S-2B: - Anhang zum Flughandbuch Nr. E-662, LBA approved 13.12.2000 or later LBA approved revision- Umrüstanweisung No. E-661 Pitts S-2C- Airplane Flight Manual Supplement No. E-1304- Instructions for Continued Airworthiness E-1306- Installation Instructions E-1305 | | 16.08.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|----------------|----------------|----------------------|--|--|---|--|---|---------------|
| 10014203 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.01354 | | | AVIATION AIRCRAFT A-1 AVIATION AIRCRAFT A-1A AVIATION AIRCRAFT A-1B | Installation of propeller MTV-14-B/183-301aTake-off- and max. Continuous-power: 126 kw by 2500 min | | Anhang zum Flughandbuch Nr. E-1231 Airplane Flight Manual Supplement No E-1230Instructions for Continued Airworthiness and Installation Instructions E-1232 | | 04.11.2005 |
| 10014238 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.01402 | REV. 2 | | 1. DV20 & 2. DA20-A1 | Extension of EASA STC EASA.A.S.01402Installation of the 2-blade MTV-21-A/170-05 | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS-VLA, Amendment 1 (not required by 21A.101, elected to comply by theapplicant)This c (...) | Aircraft Flight Manual Supplement, Doc. No. E-1909, rev 0, 14.09.2010- Instructions for Continued Airworthiness, Doc.No. E-1911, rev. 0,14.09.2010or later revisions of the above listed documents approved by EASA | none | 04.11.2010 |
| 10014239 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.01403 | | | CESSNA 182Q CESSNA 182R CESSNA F182-Q | Installation of propeller MTV-9-D/203-52 | | Anhang zum Flughandbuch Nr. E-1242Airplane Flight Manual Supplement No. E-1241Instructions for Continued Airworthiness and Installation Instructions E-1243 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 18.11.2005 |
| 10014240 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.01404 | | | DIAMOND DA-20-C1 | Installation of Propeller MT 175 R 150-2Ca | | Airplane Flight Manual Supplement No E-1255Anhang zum Flughandbuch Nr 1256Installation Instructions E-1257Instructions for Continued Airworthiness E-1281 | | 14.03.2006 |
| 10014241 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.01406 | | | 1. CESSNA 182 CESSNA 182A CESSNA 182B CESSNA 182C CESSNA 182D CESSNA 182E CESSNA 182F CESSNA 182G CESSNA 182H CESSNA 182J CESSNA 182K CESSNA 182L CESSNA 182M CESSNA 182N CESSNA 182P CESSNA F182-P | Installation of propeller MTV-9D/203-52 | | - Airplane Flight Manual Supplement No. E-1316- Anhang zum Flughandbuch Nr. 1319- Installation Instructions E-1317- Instructions for Continued Airworthiness E-1318 | | 08.06.2006 |
| 10014432 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.02480 | | | PIPER PA-42-1000 | Installation of propeller MTV-27-1-E-C-F-R(G)/CFRL250-103 (right),MTV-27-1-E-C-F-R(G)/CFR250-103 (left) | | Airplane Flight Manual Supplement No. E-1410Installation Instructions E-1411Instructions for Continued Airworthiness E-1412 | -- | 20.12.2006 |
| 10014441 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.02491 | | | CIRRUS SR22 | Installation of 3-blade MTV-9-D/198-52 constant speed Propeller | | E-1320, Rev. 1MT-Propeller AFM-S Doc. No E-1322, Original issue, approved 06. July 2006, or later EASA approved RevisionMT-Propeller Instructions for ContAW, Doc. No. E-1324, Original issue, or later approved Revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 10.07.2006 |
| 10014445 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.02498 | | | VULCANAIR P68 VULCANAIR P68 B VULCANAIR P68 C VULCANAIR P68 C-TC VULCANAIR P68 OBSERVER VULCANAIR P68 OBSERVER 2 VULCANAIR P68TC OBSERVER | P.68 "Victor", P.68B "Victor", P.68C, P.68 "Observer", P.68"Observer 2".Installation of propeller MTV-12-B-C-F/CF183-59bP.68C-TC, P.68TC "Observer".Installation of propeller MTV-12-B-C-F/CF188-53 | | P.68 "Victor", P.68B "Victor", P.68C, P.68 "Observer", P.68"Observer 2".-Airplane Flight Manual Supplement No. E-1370-Installation Instructions E-1371-Instructions for Continued Airworthiness E-1372P.68C-TC, P.68TC "Observer".-Airplane Flight Manual (...) | -- | 07.11.2006 |
| 10014491 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.02553 | | | PIPER PA-34-200T PIPER PA-34-220T | Installation of Propeller MTV-12-D-C-F/CF(L)188-53 | | Airplane Flight Manual Supplement No E-1274Installation Instructions E-1275Instructions for Continued Airworthiness E-1277 | | 13.03.2006 |
| 10014492 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.02554 | | | BEECH 60 BEECH A60 BEECH B60 | Installation of 4-blade MTV-14-B-C-F/CF188-53a full feathering constant speed Propellers | | E-1332MT-Propeller AFM-S Doc. No E-1358, Original issue, approved 06. July 2006, or later EASA approved RevisionMT-Propeller Instructions for ContAW, Doc. No. E-1360, Original issue, or later approved Revision | Applicable toDuke 60 and A60 S/N P-3 thru P-597 with Lycoming TIO-541-E1C4 engineDuke B60 S/N P-247 and upDuke B60 S/N P-247 thru P-597 with LBA STC 0662/2040 and 0666/2040all equipped with #System Gomoziq# Silencer B60-606500 For all serial numbers. | 10.07.2006 |
| 10014493 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.02555 | | | DIAMOND H-36 DIMONA | Installation of 2-blade MTV-1-A/L 160-03 constant speed propeller | | Umrüst- und Wartungsanweisung E-1433, including Anhang zum Flughandbuch E-1432 | | 12.12.2006 |
| 10014621 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.02755 | REV 3 | | DORNIER DO 228-100 DORNIER DO 228-101 DORNIER DO 228-200 DORNIER DO 228-201 DORNIER DO 228-202 DORNIER DO 228-202K DORNIER DO 228-212 | - Installation of 5-blade MTV-27-1-E-C-F-R(G) / CFRL250-55b Full Feathering and Reversing Propellers.Revision 1: Extension of STC EASA.A.S.02755 to airplane model Do 228-212.Revision 2 and 3 issued to correct some typos. | | - Doc. No. E-1436 #Installation Instructions#- MT-Propeller Instructions for ContAW, Doc. No. E-1437, Revision 1,or later approved Revision- EASA approved Airplane Flight Manual Supplement, Doc. No. E-1435, Revision 2, November 11, 2008 or later approved Revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 04.02.2009 |
| 10014837 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.03054 | | | MOONEY M20E MOONEY M20F MOONEY M20J | Installation of propeller MTV-12-B/180-59b. | | Airplane Flight Manual Supplement No. E-1469.Installation Instructions E-1470.Instructions to Continued Airworthiness E-1471. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and #Associated Technical Documentation#. Compatibility with other aircraft/engine configurations shall be determined by theinstaller. | 06.03.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|------------------|----------------|----------------------|--|---|---|---|--|---------------|
| 10014886 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.03114 | | | CESSNA 421 | Installation of 4-blade MT-Propeller Model MTV-16-1-E-C-F/CF225-37a propellers on Cessna 421 C. | | Installation Instructions Document E-1585MT-Propeller AFM-SDoc. No. E-1584, Original issue from 23. November 2007, or later EASA approved Revision.MT-Propeller Instructions for ContAW. Doc No. E-1386, Original issue from 23. November 2007, or later approved Revision. | Compliance has also been shown for all combinations (and/or) of following FAA STCs: SA438NW, SA5193NM, SA8352SW, SA00015SE, SA5811SW | 18.12.2007 |
| 10014953 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.03217 | | | PILATUS PC-12 | 5-blade MTV-27-1-N-C-F-R(P)/CFR260-65a Propeller on PilatusPC-12. | | z Airplane Flight Manual Supplement (Doc. No. E-1623), Original Issue (11 April 2008) or later approved; Master Document List (Doc. No. E-1631), Rev. 1 (26 February 2009) or later approved; Instructions for Continued Airworthiness(Doc. No. E (...)) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 16.03.2009 |
| 10014993 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.03290 | | | CESSNA 340 CESSNA 340A | z Installation Instruction Document E-1530z MT-Propeller AFM-S Doc. No E-1529, Original issue from 09. July 2007, or later EASA approved Revision; MT-Propeller Instructions for ContAW. Doc. No. E-1360, Original issue from 09. July 2007, or later approved Revision | | Installation Instruction Document E-1530MT-Propeller AFM-S Doc. No E-1529, Original issue from 09. July 2007, or later EASA approved RevisionMT-Propeller Instructions for ContAW. Doc. No. E-1531, Original issue from 09. July 2007, or later approved Revision | 1. FAA STC SA09574SC must be installed | 06.09.2007 |
| 10015073 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.03403 | | | AVIAT HUSKY A-1A AVIAT HUSKY A-1B | Installation of Constant Speed 2 Blade propeller MTV-15-B/210-58.This STC is based on LBA STC SA 1314, issued 19.03.2003. | | Flight Manual Supplement "Anhang zum Flughandbuch", Doc. No. E-891, Rev. 1 dated 30.11.2007 or later approved revision.Installation Instruction "Umrüstungsanweisung", Doc. No. E-892, Rev. 1, dated 23.11.2007 or later approved revision.Operation and (...) | | 19.12.2007 |
| 10015130 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.03481 | | | PAC 750XL | Installation Propeller MTV-16-1-E-C-F-R(P)/CFR250-55a on PAC 750XL) | | AFM Supplement, Doc. No. E-1641 dated 16.06.2008Instructions for Continued Airworthiness, Doc. No. E-1643 dated 16.06.2008. | | 22.07.2008 |
| 10015203 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.03603 | | | M7 AEROSPACE SA26-AT | Installation of 5-blade MT-Propeller Model MTV-27-1-E-C-F-R(G)/CFR210-58a propellers. | | According to MT-Propeller Master Drawing List Doc. No. E-1677 and MT-Propeller Installation Instruction Doc. No. E-1665, original issue, or later RevisionsTo be operated in accordance with MT-Propeller AFM-S Document No. E-1664, original issue, or (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 02.10.2008 |
| 10015204 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.03604 | | | PIPER PA-42 PIPER PA-42-720 | Installation of 4-blade MT-Propeller Model MTV-16-1-E-C-F-R(P)/CFR230-37b propellers. | | According to MT-Propeller Master Document List Doc. No. E-1682 and MT-Propeller Installation Instruction Doc. No. E-1671, original issue, or later RevisionsTo be operated in accordance with MT-Propeller AFM-S Document No. E-1670, original issue, or (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 02.10.2008 |
| 10015216 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.03631 | REV. 3 | | 1. SA227-AT 1. SA227-TT, SA226-T(B) 2. SA227-AC, SA227-BC 3. SA227-CC, SA227-DC | Installation of 5-blade MT-Propellers ModelMTV-27-1-E-C-F-R(G)/CFR1250-103 | The Certification Basis for the original product remains applicable to this certificate/ approval.This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certificated noise and/or emissions levels. | For Model SA226-T(B), SA227-TT:According to MT-Propeller Master Operation Manual List Doc. No. E-2084and MT-Propeller Installation Instructions Doc. No. E-2085, or laterRevisionsTo be operated in accordance with MT-Propeller AFM-S DocumentNo.E-2069, (...) | Model SA227-TT and SA226-T(B) limited to "Normal Category" with MTOW 12500 lbsAccording limitation section of AFM-S | 09.12.2011 |
| 10026785 | MT-PROPELLER ENTWICKLUNG GmbH | P-EASA.A.S.03902 | | | 95-C55, D55, E55, 58, G58 | Installation of 4-blade MTV-14-D-C-F-C/F 188 30g propellers on HawkerBeechcraft Model 95-C55, D55, E55, 58, G58. | FAR 23 including amendment 34 | Airplane Flight Manual Supplement E-1789, Issue 09-07-2009 or later approved revision.Instructions for Continued Airworthiness E-1791, Issue 10-07-2009 or later approved revision.Installation Instructions E-1790, Issue 09-07-2009. | None. | 10.08.2009 |
| 10027008 | MT-PROPELLER ENTWICKLUNG GmbH | P-EASA.A.S.02941 | | | PA-46-310P (MALIBU) PA-46-350P (MALIBU MIRAGE) | Installation of 4 blade MTV-16-1-E-C-F-R(P)/CFR206-58a on airplanePA-46-310P and PA-46-350P modified with Pratt & Whitney PT6A-21, -34 or -35 engine per FAA STC ST00541SE. | FAR 23 up to and including amendment 25 plus FAR 23.1529 at amendment26. | - Airplane Flight Manual Supplement E-1802, Issue 04, August 2009 or later approved revision- Instruction for Continued Airworthiness E-1804, Issue 04, August 2009 or later approved revision- Installation Instruction E-1803, Issue 04, August 2009 | Prerequisite for the installation of this STC is the installation of aPratt & Whitney PT6A-21, -34 or -35 engine per FAA STC ST00541SE. | 28.08.2009 |
| 10027439 | MT-PROPELLER ENTWICKLUNG GmbH | P-EASA.A.S.03889 | | | BAE JETSTREAM 3100, 3200 | Installation of 5-blade MTV-27-1-E-C-F-R(G)/CFR1250-103) propellers | FAR 23, Amndt 1 through 34The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approvalThis certificate/ approval involves a change to the certificated noiseand/ or emissions levels | -According to MT-Propeller Master Document List Doc. No. E-1764, Rev 1and MT-Propeller Installation Instructions Doc. No. E-1766, Rev 1,-To be operated in accordance with MT-Propeller AFM-S Document No.E-1765, Original Issue, or later approved Revis (...) | None | 30.09.2009 |
| 10028096 | MT-PROPELLER ENTWICKLUNG GmbH | P-EASA.A.S.03901 | | | MX-7-180 B, MX-7-180 C, MX-7-180, MX-7-180 A, MXT-7-180, MXT-7-180A | Installation of a 2-blade MTV-15-B/203-58 propeller on Maule M-4 ModelsMX-7-180, MX-7-180A, MX-7-180B, MX-7-180C, MXT-7-180, and MX-7-180A | FAR-23The Certification Basis for the original product remains applicable to this certificate/ approvalThis certificate/ approval involves a change to the certificated noiseand/ or emissions levels | -Airplane Flight Manual Supplement E-1780, Original Issue dated 09.September 2009 or later approved revision-Instruction for Continued Airworthiness E-1782, Original Issue dated09. September 2009 or later approved revision-Installation Instruction E (...) | None | 01.12.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|------------------------------|----------------|----------------------|---|--|---|---|--|---------------|
| 10023676 | MT-PROPELLER ENTWICKLUNG GmbH | | | | 206H, T206H | Installation of 5-blade MTV-5-1-D-C-F-R(A)/CFR210-56 propeller on Cessna Model 206H, T206H equipped with Rolls-Royce 250-B17F2 series engine per FAA STC No. SA01878SE | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 23, Amendment 5. This certificate/ approval involves a change to the certificated noise and/or emissions levels | Airplane Flight Manual Supplement, Doc. No. E-1830, Original Issue dated 01.12.2009-Installation Instructions, Doc. No. E-1831, Original Issue dated 01.12.2009-Instructions for Continued Airworthiness, Doc. No. E-1832, Original Issue dated 01.12.2009 or (...) | Prior to installation of this STC modification i.a.w. EASA STC 10027209 (Soloy Mark II Turbine Conversion) is required | 12.04.2010 |
| 10030966 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.00749 | REV. 1 | | CESSNA U 206G, TU 206G | Installation of Propeller MTV-5-1-E-C-F-R(M)/CFR220-54 or MTV-5-1-E-C-F/CF220-54 | | MTV-5-1-E-C-F-R(M)/CFR220-54;-Instructions for Continued Airworthiness and Installation Instructions No E-1031-Airplane Flight Manual Supplement No E-1032 MTV-5-1-E-C-F/CF220-54;-Instructions for Continued Airworthiness and Installation Instructions No (...) | 1. This STC is applicable to Cessna U 206G, TU 206G both equipped with R/R Allison 250-C20S (Soloy Turbine PAC) in accordance with LBA EMZ(STC) 0082/673a only. 2. This STC is approved only for the product configuration as defined in the approved design (...) | 19.07.2010 |
| 10031167 | MT-PROPELLER ENTWICKLUNG GmbH | | | | PC-6/B1-H2, PC-6/B2-H2, PC-6/B2-H4 | Installation of 4-blade MTV-16-1-E-C-F-R(P)/CFR240-55() propeller on Pilatus PC-6/B1-H2, PC-6/B2-H2, PC-6/B2-H4 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 23, Amendment 1-34 (not required by 21A.101, elected to comply by the applicant) Th (...) | Airplane Flight Manual Supplement, Doc. No. E-1861, dated 17.03.2010-Airplane Flight Manual Supplement (propeller speed reduction), Doc. No. E-1861-1, dated 17.03.2010-Instructions for Continued Airworthiness, Doc. No. E-1863, Revision 0, dated 17.03.2010 (...) | None | 29.07.2010 |
| 10032680 | MT-PROPELLER ENTWICKLUNG GmbH | | REV. 1 | | PA-31P | Installation of two 4-blade MTV-16-1-E-C-F/CFL225-37a propellers on Piper PA-31P Change of Certification Basis. | The Certification Basis for the original product remains applicable to this certificate/ approval: CAR 3 Amdt 1 through 8 as defined in FAA TCDS A8EAThis certificate/ approval involves a change to the requirements for environmental protection and/or a c (...) | According to MT-Propeller Master Operation Manual List Doc.No. E-1951, Rev 0 and MT-Propeller Installation Instructions Doc. No. E-1953, Rev 0 or later revisions. To be operated in accordance with MT-Propeller AFM Supplement Document No. E-1952-1 and Pil (...) | According to limitation section of AFM-S and POH-S. | 10.08.2011 |
| 10033038 | MT-PROPELLER ENTWICKLUNG GmbH | | | | TBM 700 A, TBM 700 B, TBM 700 C1, TBM 700 C2, TBM 700 N | Installation of 5-blade MTV-27-1-E-C-F-R(P)/CFR225-55f propeller | The Certification Basis for the original product remains applicable to this certificate/ approval: FAR 23 as defined in EASA TCDS A.010 This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certificated noise and/or emissions levels | According to MT-Propeller Master Operation Manual List Doc.No. E-1877, Rev 0 and MT-Propeller Installation Instructions Doc. No. E-1879, Rev 0 or later revisions. To be operated in accordance with MT-Propeller AFM Supplement Document No. E-1878, Rev 0, ap (...) | according to limitation section of AFM-S | 15.12.2010 |
| 10034434 | MT-PROPELLER ENTWICKLUNG GmbH | LBA STC (EMZ) SA0786 ISSUE 3 | REV. 2 | | PA-46-310P | 1) Installation of MTV-14-D/190-17 on PA-46-310P2) Installation of MTV-14-D/195-30a on PA-46-310P3) Installation of MTV-14-D/195-30a on PA-46-310P equipped with TCM/TSIO 550-C i.a.w. FAA STC SA00380AT4) Installation of MTV-14-D/195-30b on PA-46-310P5 (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 23 including Amendment 34 (not required by 21A.101, elected to comply by the applica (...) | For installation 1 and 2: - "Anhang zum Flughandbuch Nr. E-512", Issue 30.09.1996- "Umrüstanweisung No E-511" Issue 30.09.1996For installation 3: - "Anhang zum Flughandbuch Nr. E-566", Issue 22.09.1997- "Umrüstanweisung No. E-565", Issue 22.09.1997For (...) | None | 25.05.2011 |
| 10034435 | MT-PROPELLER ENTWICKLUNG GmbH | LBA STC (EMZ) SA0626 ISSUE 4 | REV. 1 | | PA-46-350P, PA-46R-350T | 1) Installation of MTV-14-B/190-172) Installation of MTV-14-B/195-30a3) Installation of MTV-14-B/195-30bFor installation 3 optional reduction of take-off propeller speed to 2400 rpm is included | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 23 including Amendment 34 (not required by 21A.101, elected to comply by the applica (...) | For installation 1: - "Anhang zum Flughandbuch Nr. E-466", Issue 01.06.1995- Umrüstanweisung No. E-467" Issue 01.06.1995For installation 1 and 2: - "Anhang zum Flughandbuch Nr. E-466", Issue 20.02.1998- "Umrüstanweisung No. E-467", Issue 20.02.1998For (...) | None | 07.04.2011 |
| 10036196 | MT-PROPELLER ENTWICKLUNG GmbH | | | | 200,200C,200CT,200T,A200 (12A) A200 (C-12C),A200CT (C-12D) A200CT (C-12F),A200CT (RC-12D) A200CT (FWC-12D) A200CT (RC-12G) A200CT (RC-12H),B200,B200C B200CGT,B200CT,B200GT,B200T | Installation of 5 blade MTV-27-1-E-C-F-R(P)/CFR225-55f propellers | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: 14 CFR Part 23 Amdt. 1 thru Amdt. 34 (elected to comply by the applicant) FAR 25.929 Inl (...) | E-2041; Airplane Flight Manual Supplement, revision 0, dated 21.06.2011E-2042: Installation Instruction, Revision 0, dated 21.06.2011E-2043: Instructions for Continued Airworthiness, Revision 0, dated 21.06.2011 or later revisions of the above listed documents approved by EASA | none | 19.08.2011 |
| 10038174 | MT-PROPELLER ENTWICKLUNG GmbH | | REV. 2 | | PA-31T, PA-31T1, PA-31T2 PA-31T*, PA-31T1*, PA-31T2* -135A ENGINES PER FAA STC (*) MODIFIED WITH PT6A-135 OR SA00863SE | Installation of 5-blade MTV-27-1-E-C-F-R(P)/CFR210-58d propellers | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 23, Amendment 34 for the following paragraphs: 23.33, 23.45, 23.49, 23.51, 23.53, 2 (...) | Airplane Flight Manual Supplement E-1932, Revision 0, 21.09.2011-Installation Instruction E-1933, Revision 0, 21.09.2011-Instructions for Continued Airworthiness E-1934, Revision 0, 21.09.2011-Master Operation Manual List E-1931, Revision 0, 21.09.20 (...) | None | 09.01.2013 |
| 10038512 | MT-PROPELLER ENTWICKLUNG GmbH | | REV. 1 | | CESSNA 425 | Installation of 5-blade MTV-27-1-E-C-F-R(P)/CFR210-58d propellers. Cessna 425 aircraft and models modified by EASA STC EASA.I.M.A.02944 (FAASAS786SW) to install PT6A-135 (A) engine(s). | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 23.1529 Amendment 26 This certificate/ approval involves a change to the requirements (...) | Airplane Flight Manual Supplement E-2105, Revision 0, 16.01.2012Instructions for Continued Airworthiness E-2107, Revision 0, 16.01.2012Installation Instructions E-2106, Revision 0, 16.01.2012Master Operation Manual List E-2104, Revision 0, 16.01.2012 (...) | None. | 02.03.2012 |
| 10038566 | MT-PROPELLER ENTWICKLUNG GmbH | | REV. 1 | | SR22, SR22T | Installation of 4-blade MTV-14-D/195-30b propellers. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 23, Original Issue This certificate/ approval involves a change to the requirements (...) | POH / AFM Supplement SR22, E-2128, Revision 0 (<->) 04.10.2011 POH / AFM Supplement SR22T, E-2124, Revision 0, 04.10.2011 Instructions for Continued Airworthiness E-2125, Revision 0, 04.10.2011 Installation Instruction E-2126, Revision 0, 04.10.2011 Ma (...) | None. | 02.03.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|----------------|----------------|----------------------|---|---|---|--|--|---------------|
| 10038720 | MT-PROPELLER ENTWICKLUNG GmbH | | REV. 1 | LBA STC SA 1063 | P210N, P210R T210M, T210N, T210R | This EASA STC is a revision of LBA SSTD (EMZ) SA1063, dated 21.09.1999. Installation of Propeller MTV-14-D/195-30b is included in the scope of the STC in addition to the already approved -30a blades. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. FAR 23 up to and including Amendment 34 (selected by the applicant). This certificate/ ap (...) | - Airplane Flight Manual Supplement "Anhang zum Flughandbuch Nr. E-643" for Cessna Models P210N, P210R, T210M, T210N, T210R equipped with MT-Propeller MTV-14-D/195-30a. LBA approved 21.09.1999. Airplane Flight Manual Supplement E-2024 for Cessna Models P (...) | None | 15.03.2012 |
| 10039107 | MT-PROPELLER ENTWICKLUNG GmbH | | | | TAIFUN 17 E & E II | Installation of a 2-blade MTV-1-A/L165-06 propeller on powered sailplane Taifun 17 E and Taifun E II with Limbach L2400 EB1.AB or L2400 EF1/DF1 engine | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certified noise and/or emissions levels. | - Anhang zum Flughandbuch, Nr. E-2160- Airplane Flight Manual Supplement No. E-2159-Betriebs- und Einbauanweisung, MT-Propeller, Nr. E-118-Master Document List, Doc. No. E-2161 or later revisions of the above listed documents approved by EASA | None | 10.04.2012 |
| 10039667 | MT-PROPELLER ENTWICKLUNG GmbH | | | | CESSNA 208, 208B | Installation of 5-blade MT-Propeller Model MTV-27-1-E-C-F-R(P)/CFR245-55a | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certified noise and/or emissions levels. | According to MT-Propeller Master Document List No. E-2110 and MT-Propeller Installation Instructions Doc. No. E-2095, or later Revisions. To be operated in accordance with MT-Propeller AFM-S Document No. E-2094, or later approved Revisions. To be maintained i.a.w. (...) | According to limitation section of AFM-S | 16.05.2012 |
| 10042832 | MT-PROPELLER ENTWICKLUNG GmbH | | | | M-6-235, M-7-235, M-7-235B MX-7-235, MT-7-235 | Installation of 3-blade MT-Propeller Model MTV-9-B/198-52 | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certified noise and/or emissions (...) | According to MT-Propeller Master Document List No. E-2196 and MT-Propeller Installation Instructions Doc. No. E-2193, or later Revisions. To be operated in accordance with MT-Propeller AFM-S Document No. E-2192. To be maintained i.a.w. MT-Propeller ICA Doc (...) | according to limitation section of AFM-S | 20.12.2012 |
| 10042833 | MT-PROPELLER ENTWICKLUNG GmbH | | | | M20R, M20S | Installation of 4-blade MT-Propeller Model MTV-14-D/188-30b | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certified noise and/or emissions (...) | According to MT-Propeller Master Document List No. E-2182 and MT-Propeller Installation Instructions Doc. No. E-2179, or later Revisions. To be operated in accordance with MT-Propeller AFM-S Document No. E-2178. To be maintained i.a.w. MT-Propeller ICA Doc (...) | according to limitation section of AFM-S | 20.12.2012 |
| 10042834 | MT-PROPELLER ENTWICKLUNG GmbH | | | | LC41-550FG | Installation of 4-blade MT-Propeller Model MTV-14-D/195-30b | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certified noise and/or emissions (...) | According to MT-Propeller Master Document List No. E-2206 and MT-Propeller Installation Instructions Doc. No. E-2203, or later Revisions. To be operated in accordance with MT-Propeller AFM-S Document No. E-2202. To be maintained i.a.w. MT-Propeller ICA Doc (...) | | 20.12.2012 |
| 10043106 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.02493 | | | CESSNA 414A | Installation of 4-blade MTV-14-D-C-F/CF 188-30g full feathering constant speed propeller. | | - AFM Supplement Doc. No. E-1758, Original Issue, dated 30 April 2009 or later revisions. Installation Instruction Doc. No. E-1759, Original Issue, dated 30 April 2009 or later revisions. Instructions for Continued Airworthiness (ICA) Doc. No. E-1760, Ori (...) | None. This Certificate shall remain valid unless otherwise surrendered or revoked. | 21.07.2009 |
| 10043263 | MT-PROPELLER ENTWICKLUNG GmbH | | | | G 115E | Installation of 3-blade MT-Propeller Model MTV-12-B-C/183-17e | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certified noise and/or emissions (...) | According to MT-Propeller Master Document List No. E-2266 and MT-Propeller Installation Instructions Doc. No. E-2264, or later Revisions. To be operated in accordance with MT-Propeller AFM-S Document No. E-2263. To be maintained i.a.w. MT-Propeller ICA Doc (...) | according to limitation section of AFM-S | 17.01.2013 |
| 10043312 | MT-PROPELLER ENTWICKLUNG GmbH | EASA.A.S.02493 | | | CESSNA 414A | Operational Upgrade | | - AFM Supplement Doc. No. E-1758, Original Issue, dated 30 April 2009 or later revisions. Installation Instruction Doc. No. E-1759, Original Issue, dated 30 April 2009 or later revisions. Instructions for Continued Airworthiness (ICA) Doc. No. E-1760, Ori (...) | None. This Certificate shall remain valid unless otherwise surrendered or revoked. | 21.07.2009 |
| 10043855 | MT-PROPELLER ENTWICKLUNG GmbH | | | | MU-2B-20, -25, -26, 26A, -30, MU-2B-35, -36, -36A, 40, -60 | Installation of 5-blade MT-Propeller Model MTV-27-1-E-C-F-R(G)/CFR(.....) on MU-2B. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The following paragraph(s) at a later amendment FAR 23.1529 Amdt. 26CS-36 Amdt. 2 and FAR 36 Amdt. 28. This certificate/ approval involves a change to (...) | According to MT-Propeller Master Document List No. E-2252 and MT-Propeller Installation Instructions Doc. No. E-2246 and E-2306, or later Revisions. To be operated in accordance with MT-Propeller AFM-S Document No. E-2245 and E-2305. To be maintained i.a.w. (...) | According to limitation section of AFM-S. | 28.02.2013 |
| 10043964 | MT-PROPELLER ENTWICKLUNG GmbH | | | | DHC-6 SERIES 1, 100, 110 DHC-6 SERIES 200, 210 DHC-6 SERIES 300, 310, 320 DHC-6 SERIES 400 | Installation of 4-blade MT-Propeller Model MTV-16-1-E-C-F-R(P)/CFR240-550n Viking Air Limited DHC-6 | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels. | According to MT-Propeller Master Document List No. E-2291 and MT-Propeller Installation Instructions Doc. No. E-2294, or later Revisions. To be operated in accordance with MT-Propeller AFM-S Document No. E-2293. To be maintained i.a.w. MT-Propeller ICA Doc (...) | According to limitation section of AFM-S. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 08.03.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|---|---|--|--|--|---------------|
| 10043965 | MT-PROPELLER ENTWICKLUNG GmbH | | | | GA8-TC 320 | Installation of 3-blade MT-Propeller Model MTV-9-B/200-58 on GA8-TC320 | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels. | According to MT-Propeller Master Document List No. E-2217 and MT-Propeller Installation Instructions Doc. No. E-2215, later Revisions To be operated in accordance with MT-Propeller AFM-S Document No. E-2214 To be maintained i.a.w. MT-Propeller ICA Doc (...) | According to limitation section of AFM-S Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 08.03.2013 |
| 10046759 | MT-PROPELLER ENTWICKLUNG GmbH | | | | B90, E90, C90 C90A, C90GT, C90GTI | Installation of 5-blade MT-Propeller Model MTV-27-1-E-C-F-R(P)/CFR210-58d on HBC King Air B90, C90, C90A, C90GT, C90GTI, E90 | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The following paragraph(s) at a later amendment: FAR 23.1529 Amct. 26CS-36 Amct.2 and FAR 36 Amct. 28 This certificate/ approval involves a change to (...) | According to MT-Propeller Master Document List No. E-2282 and MT-Propeller Installation Instructions Doc. No. E-2276, later Revisions To be operated in accordance with MT-Propeller AFM-S Document No. E-2275 To be maintained i.a.w. MT-Propeller ICA Doc (...) | According to limitation section of AFM-S Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 18.10.2013 |
| 10045381 | N2 DEVELOPMENT Inc. | | REV. 1 | FAA ST02436LA | 737-900ER | Installation of Cabin Interior Noise Reduction System in configuration*-1# aircraft (see MDL referenced below). | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List N2002 rev. D dated 08 October 2012 Maintenance Manual Supplement N20020301 rev. 001 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision (...)) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 31.07.2013 |
| 10045954 | N2 DEVELOPMENT Inc. | | | FAA ST02466LA | 737-900ER | Installation of cabin interior noise reduction system in configuration*-3# aircraft (see MDL referenced below). | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List (MDL) N2002 revision I dated May 15th, 2013 Maintenance Manual Supplement N20020301 rev 002 dated March 27th, 2013 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subs (...)) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 31.07.2013 |
| 10015760 | NASCO AIRCRAFT BRAKE, INC. | EASA.IM.A.S.01796 | | FAA STC ST01833LA | BOEING 737-200 BOEING 737-200C BOEING 737-300 BOEING 737-400 BOEING 737-500 | Installation of brake components listed on Nasco Aircraft Brake, Inc drawing 737101, Revision , dated August 11, 2005 for use in Goodrich brake assemblies 2-1474-3, -4, -5 and -7 in accordance with Nasco Configuration Control Drawing NCC007, Revision (...) | | see description | Limited to installation on BOEING 737-200, 200C, -300, -400, -500 For use only in Goodrich brake assemblies 2-1474-3, -4, -5 and -7 Continued Airworthiness Requirements included in the Goodrich Brake Overhaul Manual 32-40-30 Compatibility of this design (...) | 31.05.2006 |
| 10016628 | NASCO AIRCRAFT BRAKE, INC. | EASA.IM.A.S.03063 | | FAA STC ST02159LA | BOEING 767-200 SERIES | STC Validation - FAA STC ST02159LA | | NASCO Configuration Control Report, Number NCC 008, Revision NC, 13th August 2008. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 13.11.2008 |
| 10029105 | NASCO AIRCRAFT BRAKE, INC. | | | FAA STC ST02664LA | 747-100 SERIES, 747-200B SERIES, 747-200C SERIES, 747-200F SERIES, 747-300 SERIES | Installation and Replacement of Steel Brake Components in accordance with NASCO's Configuration Control Drawing, Report Number NCC 009, Revision A, dated December 2, 2009, or later FAA approved revision. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | NASCO Components Maintenance Manual 747001 shall be used with these brakes or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 05.03.2010 |
| 10033397 | NASCO AIRCRAFT BRAKE, INC. | | | FAA STC ST02321LA | DC-9-81 (MD-81) DC-9-82 (MD-82) DC-9-83 (MD-83) DC-9-87 (MD-87) MD-88 | FAA STC No. ST02321LA Validation "Installation and replacement of brake components" | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. For parts not changed or not affected by the change refer to FAA TCDS A61WE; for parts ch (...) | Nasco MD80 Compliance check list FAA project ST13341LA-T; Nasco Configuration Control Drawing-Record of drawing changes Doc. NCC010/Revision NC, dated 30 September 2010, Nasco Component Maintenance Manual containing Instruction for Continued Airworthiness (...) | Refer to FAA STC ST02321LA | 18.01.2011 |
| 10015748 | NATIONAL AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01781 | | FAA STC ST01883CH | AIRBUS A300 MODEL B2-1A AIRBUS A300 MODEL B2-1C AIRBUS A300 MODEL B2-203 AIRBUS A300 MODEL B2K-3C AIRBUS A300 MODEL B4-103 AIRBUS A300 MODEL B4-2C AIRBUS A310 AIRBUS A310-222 AIRBUS A310-304 B4-603 B4-620 B4-605R B4-622R F4-605R F4-622R C4-605 VAR.F | EASA validation of FAA STC ST01883CH Reinforced Flightdeck Door and Venting System Installation of reinforced flightdeck door and venting system defined in Drawing List FDS-DL16, Revision M, dated August 26, 2003, or later FAA approved revisions, in a (...) | | The National Aircraft Service, Inc Airplane Flight Manual Supplement Number FDS-19-AFM, FAA approved on September 15, 2003, or later EASA Approved Revisions, is required with this STC. | 1. This flightdeck door modification has not been shown to provide smoke penetration and evacuation protection as required by 14 CFR 25.831. An EASA approved smoke barrier/curtain, installed behind the flightdeck bulkhead and in front of the cargo (...) | 10.04.2006 |
| 10043676 | NATIONAL AIRCRAFT SERVICES, INC. | | | FAA STC ST02646CH | 747-100, 747-200B, 747-200F 747-200C, 747-300, 747-400 747-400F | NASI Pack Flow Control System. Modification of the Pneumatic Air Cycle Systems by a National Aircraft Service, Inc B747 Flow Control System. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List, NK-MDL24, Revision NC Airplane Flight Manual Supplement, NK24AFMS, Revision DICA; Maintenance Manual Supplement, NK-747-AM24, Revision AEWIS Compliance Plan NK-ER24.7, Revision A or later revisions of the above listed documents appro (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 18.02.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|----------------|----------------|----------------------|--|--|--|---|---|---------------|
| 10036680 | NAYAK AIRCRAFT SERVICE GmbH | | | | PA-46-310P, -350P, -500TP, PA-46R-350T | Installation of Tactical Weather Detection System (Stormscope) AvidyneTWX670. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual-Supplements: TWX670-NY-02180-08, Rev. 1. Engineering Order: TWX670-NY-02180-05, Rev. 0. Maintenance Procedures: TWX670-NY-02180-07, Rev. 0. For later revisions of the above listed documents approved by EASA. | None. | 28.09.2011 |
| 10036936 | NAYAK AIRCRAFT SERVICE GmbH | | | | F27 MARK 050 | Cabin divider for installation in Fokker 50. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | INSTALL CABIN DEVIDER IN FOKKER 50 ACCORDING TO EO F50-NY-12177-05.- INCORPORATE MAINTENANCE INSTRUCTIONS ACCORDING TO EO F50-NY-12177-07. | None | 18.10.2011 |
| 10038427 | NAYAK AIRCRAFT SERVICE GmbH | | | | PA-46-310P, PA-46-350P, PA-46-500TP, PA-46R350T | Installation of HID Xenon LDG/TAXI and LED NAV Light according NAYAK PA46-NY-09199 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | NAYAK PA46-NY-09199-MDL to be operated iaw. NAYAK AFM-S PA46-NY-09199-08 to be maintained iaw. NAYAK MM-P PA46-NY-09199-07 or later revisions of the above listed documents approved by EASA | as referred in AFM-S | 15.02.2012 |
| 10043852 | NAYAK AIRCRAFT SERVICE GmbH | | | | 112, 112B, 112TC, 112TCA 114, 114A, 114B, 114TC | Installation of HID Xenon LDG/TAXI and LED NAV Lights. Replacement of existing installation with a new HID and LED installation. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List, Nayak GmbH, document RC114-NY-10202, rev. 1, dated January 2, 2013; Instructions for Continued Airworthiness and Installation according to Nayak GmbH, document RC114-NY-10202, rev. 0, dated October 26, 2011; Aircraft Flight Manual (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 28.02.2013 |
| 10045229 | NAYAK AIRCRAFT SERVICE GmbH | | | | PA-28-140 (CHEROKEE CRUISER) PA-28-151 (CHEROKEE WARRIOR) PA-28-160 (CHEROKEE) PA-28-161 (WARRIOR II) PA-28-161 (WARRIOR III) PA-28-180 (ARCHER) PA-28-180 (CHEROKEE) PA-28-181 (ARCHER II) PA-28-181 (ARCHER III) PA-28-235 CHEROKEE PATHFINDER PA-28-236 (DAKOTA) PA-28R-180 (ARROW) PA-28R-200 (ARROW II) PA-28R-200 (ARROW) PA-28R-201 (ARROW III) PA-28R-201T (TURBO ARROW III) PA-28RT-201 (ARROW IV) PA-28RT-201T (TURBO ARROW IV) | Installation of J.P. Instruments (JP) Primary Engine Data Management System EDM930 in accordance with Engineering Order EDM930-NY03218-05. The EDM-930 is a combined electronic indicating system which simultaneously displays to the pilot powerplant and aircraft systems operating parameters. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 23, Amendment 2 for affected paragraphs: 23.21, 23.303, 23.561, 23.603, 23.773, 23.1301, 23.1305, 23.1309, 23.1321, 23.1 (...) | Airplane Flight Manual Supplement EDM930-NY-03218-08, Revision 0, 10.04.2012 or later revisions of the above listed documents approved by EASA. Master Document List, Revision 0, 11.04.2012, or later approved revisions. Installation Instruction Engineering (...) | None | 10.06.2013 |
| 10014090 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.01187 | | | FAIRCHILD SWEARINGEN SA 228 SE FAIRCHILD SWEARINGEN SA 227 SE | Installation of Thommen Air Data Display and Altimeter. | | EASA-approved Flight Manual Supplement Doc No SA226/227-NY-0212-08, Rev. 0, or later EASA-approved revision. | | 07.04.2005 |
| 10014426 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.02471 | | | CESSNA 441 SERIES | Installation of Dual Mode S Transponder GTX 330D | | Engineering Order C441-NY-0124AFM-S C441-NY-0124-08, Rev. 0, approved 14. February 2006, or later EASA approved Revision/Maintenance Procedures C441-NY-0124-07, Rev. 0, or later approved Revision | 1. Limitation according AFM-S C441-NY-0124-082. Compatibility of this modification with other previously approved modifications must be determined by the installer | 15.02.2006 |
| 10014439 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.02489 | REV. 1 | | BAE 125 -800 BAE 125 SERIES 800A BAE 125 SERIES 800B HAWKER 800 HAWKER 800 XP HAWKER 850 XP | Installation of wing mounted Xenon Taxi/Landing Light and/or Radom mounted Taxi Light | | Engineering order H800-NY-0125-05AFM-S H800-NY-0125-08, Rev 0 approved 07/04/2006 or later EASA approved revision/Maintenance Procedures H800-NY-0125-07, Rev 0 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 05.09.2006 |
| 10014440 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.02490 | | | BAE 125 -800 HAWKER 800 HAWKER 800 XP | Installation of Iridium Satcom C1000 | | Engineering Order H800-NY-0126-01AFM-S H800-NY-0126-08, Rev. 0, approved 12. May 2006, or later EASA approved Revision/Maintenance Procedures H800-NY-0125-07, Rev. 0, or later approved Revision | | 01.06.2006 |
| 10014527 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.02612 | | | BOMBARDIER CL-600-2B19 BOMBARDIER CL-600-2C10 BOMBARDIER CL-600-2D24 | Installation of RAAS compatible EGPWS Computer in CL-600-2B19-2C10-2D24 Series Aircraft | | Document Inventory List CRJ-NY-0429 Revision 1, dated 09.02.2007 or later approved revisions. Flight Manual Supplement CRJ-NY-0429-08 Revision 00 or later approved Revisions. | Certification Basis of the Change: CS 25 Amcd 2 | 28.02.2007 |
| 10014528 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.02613 | | | | Activation of RAAS function of EGPWS computer in AVRO RJ85/100 Series Aircraft | | AFM - RJ85-NY-0428-08 Rev 1 Master Document List RJ85-NY-0428-Rev 0 | As listed in the AFM | 23.02.2007 |
| 10014537 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.02624 | | | | Installation of RAAS compatible EGPWS Computer in ARO RJ85/100 Series Aircraft | | AFM - RJ85-NY-0427-08 Rev 0 Master Document List RJ85-NY-0427-08 Rev 0 | As listed in the AFM | 23.02.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|----------------|----------------|----------------------|---|---|--|--|---|---------------|
| 10014538 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.02625 | | | BOMBARDIER CL-600-2B19 BOMBARDIER CL-600-2C10 BOMBARDIER CL-600-2D24 | Activation of RAAS function of EGPWS Computer in CL-600-2B19-/2C10-/2D24 Series Aircraft | | Document Inventory List CRJ-NY-0432 Revision 1, dated 09.02.2007 or later approved revisions.Flight Manual Supplement CRJ-NY-0432-08 Revision 00 or later approved revisions | Certification Basis of the Change: CS 25 Amdt 2, additionally CRI F-01 at Issue 2 "HONEYWELL RAAS Airworthiness Approval" | 23.02.2007 |
| 10014604 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.02727 | | | DC-3C-S1C3G | Installation of emergency exit signs and floor escape path markings at DC30 series aircraft.Installation of emergency exit signs light with power source and installation of fluorescent floor path markings | | Master Document List DC3-NY-0532 Rev. 0 dated 20 Nov 2006 or later approved revision.Engineering Order DC3-NY-0532-05 Rev.0.Mechanical Installation DC3-NY-0532-03 Rev. 0.Electrical Drawings DC3-NY-0531-02 Rev. 0.Weight & Balance DC3-NY-0532-09Rev. 0 | 1. The system must be maintained in accordance with Maintenance Procedures DC3-NY-0532-07 Rev. 0 dated 20 Nov 2006 or later approved revision.2. The system must be operated in accordance with AFM supplement DC3-NY-0532-08 Rev. 0 dated 20 Nov 2006 or (...) | 05.02.2007 |
| 10014612 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.02739 | | | BEECH 1900 BEECH 200 BEECH 300 | Installation of HID Nose Gear Lights | | Installation to be in accordance with Nayak Engineering Order ref BE200/300-NY-0733-05, Rev 0 dated 12 September 2006, or later approved revision.Instructions for Continued Airworthiness as defined in Nayak Maintenance Procedures ref BE200/300-NY-0 (...) | | 12.02.2007 |
| 10014676 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.02837 | | | BOEING 737-300 | Installation of Galley G2 Varig p/n 25-30-3154 inside aircraft between STA 333.25 and STA 350.50 | | Documentation according "Übersicht der Dokumente" Rev. 3 dated 30.01.07.Compliance Check List B737-NY-0936-01A, Rev. 2 dated 30.01.07.Declaration of Completion B737-NY-0936, dated 30.01.07 | Elect to comply CS-25 ARNDT. 2 | 01.02.2007 |
| 10014753 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.02939 | | | DC-9-80 (MD-80) SERIES | Installation of flight ID capable XPDR and CTRL | | Übersicht der Dokumente No. MD80-NY-1137 dated 14.02.2007 AFMS No. MD80-NY-1137-08 dated 14.02.2007 | | 13.04.2007 |
| 10014804 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.03007 | | | CESSNA 425 | Installation of GTX 330D Mode S Transponder | | Doc. No. EO C425-NY-1240-05, R0EASA approved NAYAK AFM-S Doc. No. C425-NY-1240-08, Original Issue, or later approved RevisionNAYAK maintenance procedures. Doc. No. C425-NY-1240-07, Original issue, or later approved Revision | | 20.03.2007 |
| 10014881 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.03109 | | | MITSUBISHI MU-2B-40 MITSUBISHI MU-2B-60 | Installation of XPDR Mode-S System with Diversity, Garmin GTX 330 D (single or optional dual installation possible) | | EASA approved Compliance Check List / Compliance Statement, Doc. No. MU2-NY-0243-01A, Revision 0, dated 20-Feb-2007 or later approved revisionEASA approved Airplane Flight Manual Supplement, Doc. No. MU2-NY-0243-08, Initial Release, dated 20-Feb-2007 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 27.08.2007 |
| 10014971 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.03251 | | | FAIRCHILD SWEARINGEN SA 226 FAIRCHILD SWEARINGEN SA 227 | Installation of Honeywell MK IV EGPWS, together with MFD and Thommen Air Data Computer | | EASA approved #Übersicht der Dokumente#, Project No. SA226/227-NY-0808, Revision 0, dated 07-Nov-2007, or later approved revisionEASA approved Airplane Flight Manual Supplement, Doc. No. SA226/227-NY-0808-08, Revision 0, dated 31-May-2007, or later a (...) | | 14.11.2007 |
| 10015256 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.03709 | | | FAIRCHILD SWEARINGEN SA 226 T(| Installation of SLAR Surveillance System quick change roll | | z EASA approved #Übersicht der Dokumente#, Project No. SA226-NY-10117, Revision 1, dated 19-Feb-2009, or later approved revision, EASA approved Airplane Flight Manual Supplement, Doc. No. SA226-NY-10117-08, Revision 2, dated 22-Dec-2008, or la (...) | 1. Limited to Serial Number T2942. With SLAR equipment installed, maximum cabin seating is 4 + 1. Operator Seat must not be occupied during take-off and landing.3. Prior to installation of this modification the installer must determine that (...) | 19.02.2009 |
| 10015257 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | EASA.A.S.03710 | | | PIPER PA-34-200 PIPER PA-34-200T PIPER PA-34-220T | Replacement of existing Garmin COM/NAV and existing Garmin GPS system by Garmin GNS 530w COM/NAV/GPS system. | | - Approved Engineering Order, Doc. No. PA34-NY-10118-05, Revision 1, dated 26-Jan-2009, or later approved revision - EASA approved Airplane Flight Manual Supplement, Doc. No. PA34-NY-10118-08, Revision 2, dated 27-Jan-2009 or later approved revision (...) | 1. Limited to #Single GNS# # Installation only.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect (...) | 29.01.2009 |
| 10028781 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | | | | CESSNA 210 | Post Installation Approval and conformity check of avionics system | EASA CS-23The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | -Flight Manual Supplement: C210-NY-07136-08, Rev. 1-Instructions for Continued Airworthiness: C210-NY-07136-07, Rev. 0-Installation Manual: C210-NY-07136-05, Rev. 0 for later revisions of the above listed documents approved by EASA | Limited to S.N 57339 | 09.02.2010 |
| 10030066 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | | REV. 1 | | ATR 42-300, ATR 42-320, ATR 42-500, ATR 72-101, ATR 72-102, ATR 72-201, ATR 72-202, ATR 72-211, ATR 72-212, ATR 72-212A | Installation of single HF-9000 Communication Transceiver System in ATR42 and ATR 72This STC installs an additional HF communication device.This revision has added some ATR 42 aircraft models within the initialSTC applicability. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:Special Condition(s): H-01: Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWISCS 25 Amdt 4.25.21, 25.2 (...) | Master Document List, ref: ATR72-NY-02155 Revision 7 dated 16 April 2013-Aircraft Flight Manual Supplement, ref: ATR72-NY-02155-08 Revision 5 dated 09 August 2012-Instructions for Continued Airworthiness, ref: ATR72-NY-02155-07Revision 6 dated 20 Nov (...) | ATR Modification 2028 (Fuselage-Provide basic provisions for HF antenna strap insulator) and Mod ATR Modification 3138 (Communications-HF system installation of basic antenna) must be embodied on the aircraft at the time of this STC installation.Prior to (...) | 19.04.2013 |
| 10031264 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | | | | 1. C90A,C90GT,C90GT1 2. C90 | Installation of an Avidyne Traffic Advisory System TAS 600/610/620 | The Certification Basis of the original product and the following additional / alternative airworthiness requirements are applicable to this changeCS-23 has been elected to comply with for those areas affected by the change unless otherwise statedThe c (...) | EASA approved Master Document List, Project No. C90-NY-07137, Rev. 1, dated 03-Aug-2010, or later approved revisionEASA approved Master Document List, Project No. C90-NY-07137, Rev. 1, dated 03-Aug-2010, or later approved revisionEASA approved Flight (...) | none | 06.08.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|----------------------|----------------|----------------------|--|--|--|--|---|---------------|
| 1003154 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | | | | SA226-AT, SA226-T(B), SA226-TC SA227-AT SA227-TT, SA227-AC, SA227-DC SA26-T, SA26-AT, SA226-T, | Dual installation of GNS 430 WAAS in SA226/227 Series | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amndt. 1The requirements for environmental protection and the associated certific (...) | Instructions for Installation: GNS430-NY-03157-05, Rev. 2, dated 05.08.2010 Flight Manual Supplements: GNS430-NY-03157-08, Rev. 3, dated 05.08.2010 Instructions for Continued Airworthiness: GNS430-NY-03157-07, Rev. 1, dated 05.08.2010 or later revisions of the above listed documents approved by EASA | None | 21.12.2010 |
| 10033946 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | | | | 401, 401A, 401B, 402, 402A, 402B, 402C, 411, 411A, 414, 414A, 421, 421A, 421B, 421C, 425. | Installation of a Traffic Advisory System Avidyne TAS610/620. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 1 (not required by 21A.101, elected to comply by the applicant) The req (...) | Engineering Order ,,,,,,C421-NY-08167-05, Rev. 1 Maintenance Procedures,,,,,C421-NY-08167-07, Rev. 1 Flight Manual Supplement ,,,,,C421-NY-08167-07, Rev. 1 or later revisions of the above listed documents approved by EASA. | None | 21.02.2011 |
| 10033970 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | | REV. 2 | | 401, 401A, 401B, 402, 402A, 402B, 402C, 411, 411A, 414, 414A, 421, 421A, 421B, 421C, 425 | Single installation of Garmin GNS 430 (A)W/GNS 530 (A)W and optional installation of Roll Steering Converter GDC31. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: EASA CS-23, Amendment 1 (elected to comply by the applicant, not required by 21A.101) Th (...) | Engineering Order ,,,,,C421-NY-08168-05(->,-<)- Rev. 2 Flight Manual Supplement,,,,,C421-NY-08168-08a(->,-<)- Rev. 2 C421-NY-08168-08b, Rev. 1 Maintenance Procedures,,,,,C421-NY-08168-07, Rev. 1 or later revisions of the above listed documents approved. | None | 14.03.2011 |
| 10034483 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | | | | 560XL | Installation of HID/LED LDG/REC/TAXI light Replacement of existing conventional LDG/REC/TAXI lights by low power HID/LED lights | CS 25 amendment 10 The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Document List to project C560XL-NY-1170, Revision 3 from 07 April 2011 CA C560XL-NY-1170-07, Revision 1 from 07 April 2011 or later revisions of the above listed documents approved by EASA | None | 07.04.2011 |
| 10034925 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | | REV. 1 | | T303 | Installation of 1. Dual GPS/NAV/COMM Garmin GNS430/530(A)W2-E-HSI Sandel SN3308 with Directional Gyro Century NSD360A3. Various other NAV/COMM related Avionic Components | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Engineering Order: C303-NY-07165-05, Rev. 1 Flight Manual Supplements: C303-NY-07165-08a, Rev. 1C303-NY-07165-08b, Rev. 1C303-NY-07165-08c, Rev. 1C303-NY-07165-08d, Rev. 1C303-NY-07165-08e, Rev. 1C303-NY-07165-08f, Rev. 1-Maintenance Procedures: C303- (...) | None | 16.05.2011 |
| 10037575 | NAYAK AIRCRAFT SERVICE GmbH & Co.KG | | REV. 1 | | EMB-135BJ EMB-135ER, EMB-135LR EMB-145EP, EMB-145MP EMB-145ER, EMB-145LR EMB-145LU EMB-145MK, EMB-145EU | Installation of HID Taxi/Landing light, inspection light on EMB-145 aircraft. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Document List doc. N° EMB145-NY-04188-MDL, Rev. 2 dated 07.12.2011 Airplane Flight Manual Supplement doc. N° EMB145-NY-04188-08 Rev. 4 dated 02.12.2011 Instructions for Continued Airworthiness - Maintenance procedures doc. N° EMB145-NY-04188-07 Re (...) | Not Applicable | 09.12.2011 |
| 10016258 | NEW FLIGHT COOPERATION | EASA, I.M.A.S. 02537 | | FAA STC SA4954NM | CITATION 550-0001 TO -0800 | Gross Weight Increases based on Banson Aircraft Corp. STC LBA STC Nr. 2048, 05 June 1996; Spanish STC No. S325-1, 06 Febr. 2003. | | New Flight Corporation Drawing List Doc. 90A890, Rev. A or later approved revision; AFM Supplements FMS-972 and (metric) 2BRA Maintenance Manual Supplements MMS-921 (S/Ns 0001-0626) MMA-923 (S/Ns 0627-0800) | | 19.11.2007 |
| 10015648 | NEW SYSTEMS ENGINEERING | EASA, I.M.A.S. 01622 | | FAA STC ST09384SC | SAAB 340 SAAB 340B | Installation of a single Universal Avionics Flight Management System in SAAB SF340A and SAAB 340B FAA STC ST09384SC, Rev. 2 | | FAA STC ST09384SC, Rev. 2, including New Systems AFM Supplement 227-00-0011, Rev. D, FAA approved February 27, 2006 | None | 19.04.2006 |
| 10039579 | NEW SYSTEMS ENGINEERING | | REV. 1 | FAA STC ST10977SC | DHC-8-102/-103/-106 DHC-8-201/-202 DHC-8-301/-311/-314/-315 | Installation of Dual Universal Avionics Flight Management Systems (FMS) with Global Positioning System (GPS) FAA STC ST10977SC | FAR 25 as per EASA, I.M.A. 191, Issue 5 (incl. SC H-01) The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 25, Amendment 123 The requiremen (...) | Installation in accordance with New Systems Master Drawing List 379-00-0001, dated 02.05.2012 *) Flight Operations in accordance with New Systems Flight Manual Supplement 379-00-0004, Rev. IR, dated 01.06.2010 *) Instructions for Continuing Airworthiness (...) | Installation is limited to serial numbers listed in the latest approved version of FMS Instruction Top Drawing 379-02-0001. | 21.05.2012 |
| 10040905 | NEXTANT AEROSPACE L.L.C. | | | FAA STC ST03960AT | 400A | Installation of Rockwell Collins Venue In-Flight Entertainment (IFE), Aircell Access Cabin II Inflight Phone and High Speed Internet System with Wireless Local Area Network (WLAN) Avionics 115V AC 60 Hz Cabin Inverter, and Enteq LED Cabin Lighting System. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 25 Amendment 123 The requirements for environment (...) | Installation in accordance with Nextant Aerospace LLC Master Drawing List Doc No DB1002067 Revision E, dated January 26, 2012. Operation in accordance with Nextant Aerospace LLC FAA Approved Flight Manual Supplement: When Rockwell Collins Venue System i (...) | This STC is intended to provide internet connection and E-Mail service to the aircraft cabin using portable electronic devices (PED's). Any other intended function of this equipment will require a re-examination of the certification basis. | 02.08.2012 |
| 10040906 | NEXTANT AEROSPACE L.L.C. | | | FAA STC ST03110CH | 400A | Installation of L3 Communications Digital Flight Data Recorder. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 25 Amendments as detailed on FAA STC ref ST03110CH. The requirements for environmen (...) | Installation in accordance with Nextant Aerospace LLC Master Drawing List Doc No NX1111001 Revision 1, dated May 29, 2012; Operation in accordance with Nextant Aerospace LLC FAA Approved Flight Manual Supplement No NX1111004 Revision IR, dated May 17, (...) | None. | 02.08.2012 |
| 10042091 | NEXTANT AEROSPACE L.L.C. | | | FAA STC ST10959SC | BEECHJET 400A | Installation of Rockwell Collins Pro Line 21 Electronic Flight Instrument System (EFIS) with Rockwell Collins FMS-6100, Localiser performance with Vertical Guidance (LPV) approach capability and Universal Systems Company (UASC) Terrain awareness warning System (TAWS). | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 25 Amendments 25-1 through 25-40 FAR 25.1335, 25.1351(d), 25.1353(c)(5), 25.1447 a (...) | Installation in accordance with Nextant Aerospace LLC Master Drawing List Doc No 373-00-001 Rev W, dated March 29, 2012; Operation in accordance with: For aircraft equipped with Pratt & Whitney JT15D Engines FAA Approved Flight Manual Supplement, Nexta (...) | None. | 07.11.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|--------------------------------|--|---|--|--|---|---------------|
| 10042353 | NEXTANT AEROSPACE L.L.C. | | | FAA STC ST02371LA | 400A | Installation of Williams International FJ44-3AP engines as replacement for the original Pratt & Whitney Canada JT15D-5 units. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The FAA/EASA certification basis on TCDS A16SW was updated by ST02371LA from the above (...) | Installation of Williams International FJ44-3AP engine with Full Authority Digital Engine Control (FADEC) and associated systems, in accordance with Nextant Aerospace Master Drawing List No.: MDL-NXT-001. Revision IR dated August 22, 2011. Operation in accordance with FAA Approved Airplane Flight Manual | None. | 26.11.2012 |
| 10040787 | NORTH FLIGHT DATA SYSTEMS, L.L.C. | | | FAA STC SR09468RC | EC135 P1, EC135 P2, EC135 P2+, EC135 T1, EC135 T2, EC135 T2+ | Installation of the Outerlink Corporation CP-3 or CP-3i Automatic Flight Following System, Optional Electronic Flight Bag Provisions, and Flight Data System in accordance with North Flight Data System, LLC Master Installation List, Document N° NL4620730 (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | North Flight Data System, LLC Master Installation List, Document N° NL462073006, Revision M, dated October 10, 2011. - North Flight Data Systems, LLC Instructions for Continued Airworthiness, Document N° NM462073006, Revision E, dated October 20, 2011 (...) | none | 26.07.2012 |
| 10016294 | NORTHEAST ENGINEERING & DEVELOPMENT | EASA.IM.A.S.02591 | | TCCA LSTC NO. O-LSA05-321/D IS | PILATUS PC-12/45 | Installation of GNS 430 (COMM/NAV/GPS), GNS 530 (COM/NAV/GPS), KDR 510 (Data Link Weather Receiver), KGS SS50 (Static Inverter), XM Radio Receiver with XM Radio Display & Control Head (Cabin), and necessary antennas. | | Installation of GNS 430 (COMM/NAV/GPS), GNS 530 (COM/NAV/GPS), KDR 510 (Data Link Weather Receiver), KGS SS50 (Static Inverter), XM Radio Receiver with XM Radio Display & Control Head (Cabin), and necessary antennas. Transport Canada approved Master D (...) | This STC is limited to Pilatus PC-12/45, Serial number 659. Prior to the installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modifications will introduce (...) | 16.10.2007 |
| 10017039 | NORTHEAST ENGINEERING & DEVELOPMENT | EASA.IM.R.S.01463 | | TCCA STC SH97-59 | ROBINSON R44 ROBINSON R44 II | Bear Paw Installation. | | Master Drawing List 960501, Issue 2, dated November 4, 2004; Instructions for Continued Airworthiness 9605ICA, Issue 2, dated January 16, 2008. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 18.07.2008 |
| 10042990 | NORTHEAST ENGINEERING & DEVELOPMENT | EASA.IM.A.S.02241 | REV. 1 | | CESSNA T207A | STOL Installation. | | Installation must be in accordance with Northeast Engineering & Development Ltd. Master Drawing List MDL No. 040201 Issue 1, dated 29 June 2004 or later approved revision. - Placards must be in accordance with Northeast Engineering & Development Ltd. Dra (...) | 1. This Installation is limited to the Serial No. 20700706 (Registration C-FYBZ). 2. Compatibility of this modification with other previously approved modifications must be determined with the installer. 3. This STC is approved only for the product conf (...) | 20.12.2006 |
| 10014817 | NORTHERN AVIONICS S.R.L. | EASA.A.S.03025 | | | PIPER PA-31T | Installation of Mode S Diversity XPDR Honeywell MST-67A (Elementary Surveillance). | | Installation must be in accordance with Master Document List included in the Northern Avionics Modification Bulletin NAV-324 Rev.0 dated 1st March 2007 or later Approved Revisions. Operation must be in accordance with Northern Avionics Airplane Flight (...) | | 13.07.2007 |
| 10017108 | NORTHERN AVIONICS S.R.L. | EASA.R.S.01084 | | | AGUSTA A-109 A AGUSTA A-109 All AGUSTA A-109 C | NAV 277 "Installation of an alternate static system on board to rotorcraft A 109 A/All/C" | | Continuous Airworthiness Instruction Rotorcraft Flight Manual Supplement AFM NAV 277 Rev. 0 | | 19.01.2006 |
| 10017115 | NORTHERN AVIONICS S.R.L. | EASA.R.S.01097 | | | AGUSTA AB-412EP | Installation of an EGPWS Honeywell MK XXI. | | RFMS No. RFM NAV-229 | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.08.2005 |
| 10027831 | NORTHERN AVIONICS S.R.L. | | | | A109S | NAV-396 "Installation of Helicopter Egress Light System (HEEL)" | The Certification Basis for the original product remains applicable to this certificate. | RFM Supplement NAV-396 Rev. 0 or later revisions of the above document approved by EASA - Modification Bulletin NAV-396 Rev. 0 | None. | 09.11.2009 |
| 10035325 | NORTHERN AVIONICS S.R.L. | | REV. 1 | | AW109SP | Installation of a Relay box and Control Panel for ICS VIP application. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement for ICS VIP Miscellaneous Control Panel (KIT P/N KNAV-418), RFMS NAV 436 Rev. 0 or later revisions of the above listed documents approved by EASA. Instructions for Continuous Airworthiness ICA-NAV436, Revision 0, dat (...) | Required pre-modification: Agusta electrical/mechanical provision n°109-B810-30(-) | 05.08.2011 |
| 10035500 | NORTHERN AVIONICS S.R.L. | | | | AB206B | NAV 427 "Installation of Alternate Static System" - This kits consists in the installation of an additional line to the original static line and a 3-way valve/switch (with the air inlet port) to pick up the static pressure in the cabin when the externa (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-27 Amdt. 2 (Ref. NAV-427 Rev. 0 Sec. VIII) The requirements for environmental protec (...) | RFMS NAV 427 Rev. 0 (RFM Supplement & Supplemento al Manuale di Volo) Technical Modification Bulletin NAV-427 Rev. 0 or later revisions of the above listed documents approved by EASA | This STC is applicable only to the helicopters S/Ns from 8590 to 8758. | 27.06.2011 |
| 10035741 | NORTHERN AVIONICS S.R.L. | | REV. 1 | | AW109SP | Installation of In-flight Entertainment System (IES) (composed by kit P/N KNAV-419 and KNAV-420) Certificate rev. 1 issued due to change of Modification Bulletin/Revision level. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Modification Bulletin NAV-437, Rev. 1, dated 13 October 2011 | Must be installed in conjunction with Northern Avionics kit P/N KNAV-418 | 21.10.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|---------------------|----------------|----------------------|--|--|---|---|---|---------------|
| 10036508 | NORTHERN AVIONICS S.R.L. | | | | AW109SP | NAV 434 "Installation of Helicopter Emergency Egress Light (HEEL)" | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Modification Bulletin NAV-434 Rev. 0 | None | 14.09.2011 |
| 10036765 | NORTHERN AVIONICS S.R.L. | P-EASA.R.S.01575 | REV. 1 | | A109A, A109 All, A109C | Installation of a retractable swivelling landing light (searchlight). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement RFMS NAV-371 Rev. 0-Instructions for Continued Airworthiness ICA-NAV371R0, dated 01 July 2011 for later revisions of the above listed documents approved by EASA-Installation Instructions I-NAV371-1R0, dated July 20 (...) | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 15.04.2013 |
| 10039669 | NORTHERN AVIONICS S.R.L. | | | | AW109SP | Installation of In-flight Entertainment System (IES) (composed by KitPIN KNAV-419 and KNAV-442). | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Modification Bulletin NAV-442 Rev. 0, dated May 2012. | None. | 16.05.2012 |
| 10046079 | NORTHERN AVIONICS S.R.L. | | | | AW109SP | NAV 467 "Installation of Cabin WLAN and Web connection in the helicopter AW109SP" | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -RFM Supplement RFMS NAV-467 or later revisions of the above listed documents approved by EASA-Modification Bulletin NAV-467, Rev. 1 (including Compliance Checklist)-Ground Test Procedures (with Internet Sensor) NAV-467-005, Rev. 1-Flight Test Proceed (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 13.08.2013 |
| 10015366 | NORTHROP GRUMMAN GUIDANCE AND | EASA.IM.A.S.00759 | REV 2 | FAA STC ST01149LA | AIRBUS A300 MODEL B2-203 AIRBUS A300 MODEL B4-103 AIRBUS A300 MODEL B4-203 | Installation of Litton Heading and Attitude System (LHAS) LTR-97. This STC is the result of the FAA STC No. ST01149LA validation. | | Top installation Drawing List, Document No. 454170-14 Revision C, dated 27 June 2001. | 1. This STC is limited to the installation of three LHAS LTR-97 (triple configuration). 2. The Airplane Flight Manual Supplement, Document No. 434593-EASA dated 25 May 2005 is required as part of this installation. 3. This STC is approved only for the (...) | 31.07.2006 |
| 10015825 | NORTHROP GRUMMAN GUIDANCE AND | EASA.IM.A.S.01897 | REV 1 | FAA STC SA8027NM | | Replacement of old inertial navigation equipment with new technology | | 1. Top Installation Drawing LTN-92 INS in B747-287B, Litton Document 454065-09, Rev. N/C.2. Engineering Authorization, Litton Document 464428, Rev. N/C.3. Flight Manual Supplement to Boeing B747-200 Series Airplane, Litton Document 464427, (...) | Litton Airplane Flight Manual Supplement, Doc No 464427, N/C, FAA approved on August 8, 1994, or later approved revisions, is required as part of this modification. This STC is approved only for the Boeing 747-287 model airplanes, and it applies only (...) | 28.11.2006 |
| 10015395 | NORTHWEST AEROSPACE TECHNOLOGIES In | EASA.IM.A.S.01096 | | | BOEING 757-200 | Installation of ELS/EHS S system Upgrade | | Northwest Aerospace Technologies MDL N0212-D10 Rev A dated June 2005 or later approved revision. AFM supplement N0212-D800 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 15.08.2005 |
| 10028415 | NORTHWEST AEROSPACE TECHNOLOGIES In | P-EASA.A.S.03966 | REV. 1 | | A320-232 | Interior Reconfiguration and In Flight Entertainment (IFE) Installation on Ethihad Airways A320-232 Aircraft Revision of the EASA 10028415 to remove EWIS limitation | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- CRI H-01/Special Condition H-01 "Enhanced Airworthiness (...) | - NAT Certification Summary, ref. doc. N0340-D90, Revision -, dated 5th January 2010- NAT Master Drawing List, ref. doc. N0340-D10, Revision E, dated 1st April 2010- NAT EAPAS/FTS ICA document, ref. doc. N0340-D10-D79, Revision -, dated 2nd June 2010 or (...) | None | 05.08.2010 |
| 10029095 | NORTHWEST AEROSPACE TECHNOLOGIES In | P-EASA.IM.A.S.03043 | | FAA STC ST02052SE | B777-200 | First Class Cabin Upgrade - BA 777-200 Aircraft | EASA TCDS No. I.M.A.003 The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | -FAA STC No. ST02052SE dated 24 February 2010.-FAA SoC dated 26 February 2010.-Northwest Aerospace Technologies, Inc. (NAT) Master Drawing List N0305-D10, Revision D dated 2 February 2010 or later approved revision or later revisions of the above listed (...) | Concurrent installation of EASA STC No. 10029094 (EASA Project NoP-EASA.IM.A.S.02978) is required. Concurrent installation of EASA STC No. EASA.IM.A.S.01797 is required. This modification may not be installed after 30 August 2010 unless (1) previously (...) | 03.03.2010 |
| 10029975 | NORTHWEST AEROSPACE TECHNOLOGIES In | P-EASA.IM.A.S.03214 | | FAA STC ST10594SE-T | B747-400 | First Class Cabin Upgrade BA 747-400 aircraft | EASA TCDS No. I.M.A.196 The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | -FAA STC Number ST02063SE dated 30 April 2010.-FAA SoC dated 03 May 2010.- Northwest Aerospace Technologies, Inc. (NAT) Master Drawing List N0313-D10, Revision D dated 04 April 2010 or later approved revision or later revisions of the above listed docu (...) | Concurrent installation of EASA STC 10029964 is required. This modification may not be installed after 30 August 2010 unless (1) previously approved electrical wiring interconnect system (EWIS) instructions for continued airworthiness (ICA) have been e (...) | 07.05.2010 |
| 10030495 | NORTHWEST AEROSPACE TECHNOLOGIES In | P-EASA.A.S.03958 | | | A319-132 | Interior Reconfiguration and In Flight Entertainment (IFE) Installation on Ethihad Airways A319-132 Aircraft | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- CRI H-01/Special Condition H-01 "Enhanced Airworthiness (...) | NAT Certification Summary, ref. doc. N0338-D90, Revision -, dated 17th June 2010- NAT Master Drawing List, ref. doc. N0338-D10, Revision C, dated 16th June 2010 or later revisions of the above listed documents approved by EASA | None. | 18.06.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|--------------|----------------|-----------------------|------------|---|---|---|---|---------------|
| 10031457 | NORTHWEST AEROSPACE TECHNOLOGIES In | | | FAA STC ST02079SE | B777-300 | First and Business Class Cabin Interior - BA 777-300 aircraft | EASA TCDS No. IM.A.003The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated certified noise and/or emissions levels of the original product (...) | -FAA STC Number ST02079SE dated 10 August 2010-FAA SoC dated 17 August 2010 Northwest Aerospace Technologies Inc. (NAT) Master Drawing List N0333-D10, Revision B dated 25 July 2010 or later approved revision or later revisions of the above listed docum (...) | None | 17.08.2010 |
| 10032214 | NORTHWEST AEROSPACE TECHNOLOGIES In | | | | A320-232 | Cabin Interior Reconfiguration and In Flight Entertainment (IFE) Modifications on Ethad Airways A320-232 Aircraft | Certification Basis as per EASA TCDS A.064The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval - CS 25 Appendix H and AMC Appendix H25.5The (...) | - NAT Certification Summary, ref. doc. N0362-D90, Revision -, dated 17th October 2010- NAT Master Drawing List, ref. doc. N0362-D10, Revision B, dated 15th October 2010 or later revisions of the above listed documents approved by EASA | Not applicable | 18.10.2010 |
| 10038268 | NORTHWEST AEROSPACE TECHNOLOGIES In | | | FAA STC NO. ST02412LA | 777-200 | First and Economy Class Cabin Interior - BA 777-200 Aircraft | IM.A.003The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CRI H-01 EWIS The requirements for environmental protection and the associated (...) | FAA STC No. ST02412LA, dated 18 January 2012. Northwest Aerospace Technologies, Inc. (NAT) Master Drawing List N0366-D10, Rev "C", dated December 13, 2011 or later approved revisions. Instructions for Continued Airworthiness Document No. N0366-D70, Rev" (...) | Concurrent installation of the following STCs are required for the installation of this STC: 1) Flight Structures, Inc. EASA STC 10015720, Business Class Seat Installation. 2) Thales Avionics, Inc. EASA STC No. 10038267 (EASA Project No.0010011945), Pas (...) | 02.02.2012 |
| 10038505 | NORTHWEST AEROSPACE TECHNOLOGIES In | | | FAA STC ST02195SE | B747-400 | Installation of a IMS RAVE In-flight Entertainment System into the economy class passenger seats of a Boeing 747-400 aircraft in accordance with Northwest Aerospace Technologies Inc. Master Drawing List N0374-D10, Rev. J dated December 27, 2011 or later (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Northwest Aerospace Technologies Inc. Master Drawing List N0374-D10, Rev. J dated December 27<(-)> 2011 and documents referenced in this MDL or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None. | 22.02.2012 |
| 10039202 | NORTHWEST AEROSPACE TECHNOLOGIES In | | REV. 1 | | B747-400 | B747-400 Cabin re-configuration in accordance with Northwest Aerospace Technologies Inc. Master Drawing List N0375-D10 Rev. A dated April 12, 2012 | This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certified noise and/or emissions levels. CRI H-01 Enhanced Airworthiness Programme for Aeroplane Systems CRI D-01 Use of FAA Policy (...) | Northwest Aerospace Technologies Inc. Master Drawing List N0375-D10 Rev. A. Instructions for Continued Airworthiness - Principle Manual for the Interior Reconfiguration for Lufthansa Boeing 747-400 Aircraft N0375-D70 Rev. A or later revisions of the above listed documents approved by EASA | The use of the baby bassinet is not permitted unless a "Remove before Taxi, Take-off and Landing" placard is installed either on the baby bassinet or on the surface on which the baby bassinet is mounted. These of cabin crew procedures to show compliance (...) | 07.05.2012 |
| 10040082 | NORTHWEST AEROSPACE TECHNOLOGIES In | | | | B747-400 | Interior Reconfiguration of VAA 747-400 Aircraft: Cabin Configuration and Passenger Amenities Upgrade and Standardization in accordance with NAT Master Data List N0377-D10 Rev. B, dated May 22, 2012 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CRI H-01 Enhanced Airworthiness Programme for Aeroplane Systems The requirements for en (...) | N0377-D10 Rev B, dated May 22, 2012 and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 Enhanced Airworthiness Programme for Virgin Atlantic Airways (VAA) Boeing 747-400 Aircraft: N0377-D71 (...) | A/C serial numbers: 30885, 32337, 32338, 32339, 32340, 28757 and 29406. | 12.06.2012 |
| 10040295 | NORTHWEST AEROSPACE TECHNOLOGIES In | | | | B777-300ER | N0401: Certification of Deployment of In-Seat Video Monitors During Taxi, Take-Off and Landing - Business Class Seats on Cathay Pacific B777-300(ER) Aircraft | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CRI H-1 - ICA on EWIS | N0401-D90, Revision -, dated June 6, 2012 - Certification Summary for the Deployment of In-Seat Video Monitors During Taxi, Take-Off and Landing - Business Class Seats on Cathay Pacific B777-300(ER) Aircraft or later revisions of the above listed documents approved by EASA | Limited to 777-300ER as per EO 20N00968 | 22.06.2012 |
| 10041628 | NORTHWEST AEROSPACE TECHNOLOGIES In | | | | 777-300ER | N0419 Cabin Interior Modifications on Jet Airways Boeing 777-300ER Aircraft | The Certification Basis for the original product remains applicable to this certificate/ approval. | N0419-D90, Rev A, dtd Oct 2, 2012- Certification Summary for the Cabin Interior Modifications on Jet Airways Boeing B777-300(ER) Aircraft Drawing List N0419-D10 Rev. A dtd. Sep 28, 2012 Instructions for Continued Airworthiness N0419-D70 Rev. Adtd. Aug 3 (...) | None | 03.10.2012 |
| 10043377 | NORTHWEST AEROSPACE TECHNOLOGIES In | | REV. 2 | | 777-300 | Interior Reconfiguration for Cathay Pacific Boeing 777-300 Aircraft | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Drawing List Interior Reconfiguration Cathay Pacific Airways Boeing 777- 300 Aircraft - N0435-D10, Revision B- Airways Boeing 777-300 Aircraft Instructions for Continued Airworthiness - Principle Manual for an Interior Reconfiguration on Catha (...) | The STC is limited to the following serial numbers: 27504-27510, 33702-33704, 34243, 34244. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design (...) | 01.02.2013 |
| 10044745 | NORTHWEST AEROSPACE TECHNOLOGIES In | | | | A321-231 | Cabin Interior modifications and In Flight Entertainment installation on Dragon Air. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Certification Plan N0405-D01 Rev. B, dated 20/02/2013- Certification Summary N0405-D90 Rev. A, dated 02/05/2013- Master Drawing List N0405-D10 Rev. C, dated 30/04/2013- Instructions for Continued Airworthiness N0405-D70 Rev. B, dated 30/04/2013 or 1 (...) | The approval is limited to MSN 993, MSN 1024, MSN 633, MSN 1695, MSN 1984 and MSN 2021. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design cha (...) | 03.05.2013 |
| 10045515 | NORTHWEST AEROSPACE TECHNOLOGIES In | | REV. 1 | | A340-313 | Interior reconfiguration of Cathay Pacific A340 Aircraft Premium economy class seat installation. For additional information, refer to NAT Certification Plan N0406-D01NB : As per certification plan, the applicable aircraft Manufacturer's Serial Numbers (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification plan N0406-D01 rev B Master Drawing List : ref N0406-D10 rev B Certification Summary ref N0406-D90 rev AICA: ref N0406-D70 rev -(and EPAS/FTS ICA Document : N0406-D79 rev B) or later revisions of the above listed documents approved by EASA. | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product. | 03.07.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|----------------------|---|--|--|---|--|---------------|
| 10045832 | NORTHWEST AEROSPACE TECHNOLOGIES In | | REV. 1 | | 777-300ER | First Class Cabin Refresh of Cathay Pacific Aircraft | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | - Master Drawing List, Northwest Aerospace Technologies, Inc. document No. N0407-D10, Revision A, dated 21 July 2013. - Flight Manual Supplement, Northwest Aerospace Technologies, Inc. document No. N0407-D80, Revision A, dated 10 July 2013. - Instruction (...) | The modification approval is limited to aircraft serial numbers 36154, 35299, 34432, 36155, 36156, 36632, 35300, 35301, 36833, 36157, 36158, 36161, 36159, 36165, 36160, 36164, 36162, 36163, 39232, 37896, 39233, 37901, 39234, 37897, 42142, 42143, 41429, (...) | 22.07.2013 |
| 10015367 | NORTHWEST COMPOSITES | EASA.IM.A.S.00763 | | FAA STC ST01565SE | BOEING 737-300 BOEING 737-400 | Stowage bin Extension. | | Northwest Composites Master Drawing List Ref.: A07800 Revision C dated 8th June 2005 or later FAA approved revision. FAA STC No. ST01565SE dated 9th June 2005. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 17.06.2005 |
| 10015740 | NORTHWEST COMPOSITES | EASA.IM.A.S.01764 | | FAA STC ST00867SE | DC-9-82 DC-9-83 | Stowage Bin Extension/Modification to overhead stowage bin located in the passenger cabin in order to extend the bins for aircraft originally equipped with Heath-Techna Interiors stowage bins. | | - Master Drawing List A075000 rev. F dated 20th January 2006 or later approved revision. - Instruction for Continued Airworthiness 25-20-03 rev. 5 dated 3rd November 2005 or later approved revision. | 1. Aircraft listed on Master Drawing List A075000 rev. F dated 20th January 2006 or later approved revision. 2. This approval of this change in type design applies to the above model aircraft only. 3. This approval should not be extended to other aircraft (...) | 28.03.2006 |
| 10015741 | NORTHWEST COMPOSITES | EASA.IM.A.S.01765 | | FAA STC ST00868SE | DC-9-82 (CF6) DC-9-83 (MD-83) | Stowage Bin Extension/Modification to overhead stowage bin located in the passenger cabin in order to extend the bins for aircraft originally equipped with C&D Interiors stowage bins. FAA STC ST00868SE issue on 21st June 2001 | | - Master Drawing List A074000 rev. J dated 3rd November 2005 or later approved revision. - Instruction for Continued Airworthiness 25-20-02 rev. 5 dated 3rd November 2005 or later approved revision. | 1. Aircraft listed on Master Drawing List A074000 rev. J dated 3rd November 2005 or later approved revision. This approval of this change in type design applies to the above model aircraft only. This approval should not be extended to other aircraft (...) | 16.02.2006 |
| 10042952 | Not assigned | EASA.IM.A.S.01130 | | FAA STC SA01130SE | PA-38-112 | Installation of Tomahawk Fatigue Life Extension Wing Spar Modification. FAA STC SA1130SE. Validation is in accordance with EASA Certification Directorate Exposition Part 2, Subpart 3 Internal Working Procedure Supplemental Type Certification (STCP) Iss (...) | | -, FAA STC SA1130SE, -, Sterling Aviation Installation Instructions SA38-112-500 Revision D dated February 8 2007, -, Sterling Aviation Technologies Inc. Drawing List SA500 Revision N dated September 24 2002 or later approved revision. | 1. Approval restricted to modifications carried out in accordance with Sterling installations SA38-112-500 Revision D dated February 08 2007. Modifications completed to earlier installation instructions are not acceptable. 2. This STC is approved only (...) | 08.03.2007 |
| 10042989 | Not assigned | EASA.IM.A.S.02166 | | FAA STC SA3783NM | BEECH 200, 200C, 200CT, 200T BEECH B200, B200C, B200CT BEECH B200T, 300, 1900, 1900C | Installation of upper wing reinforcing kit. | | In accordance with FAA sealed Aviadesign Inc Master Drawing List E.O.-B-8808, dated March 10, 1988, or later approved revision. | 1. Reinforcing strap and original wing structure covered by the strap tube inspected every 1000 flying hours or annually for any signs of cracking or corrosion. 2. This STC is approved only for the product configuration as defined in the approved design (...) | 19.09.2006 |
| 10017230 | NSE INDUSTRIES S.A. | EASA.R.S.01401 | REV. 5 | | AS350B, AS350B1, AS350B2 AS350BA, AS350BB, AS350D | Installation of BRITE SAVER 2 Usage Monitoring System. Revision 1 - includes torque and memory management minor change. Revision 2 - includes addition of the possibility to install the ultrasonic air/ground switch and second source for the SUB Connectors (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The change under revision 4 complies with FAR Part 27, amends 27-1 through 27-35 require (...) | ECT Industries Rotorcraft Flight Manual Supplement RFMS-REC-ARA-AS350-EN, third issue, revision 02 dated 31 May 2012 or later revisions of the above listed documents approved by EASA. Conformity Statement and Documentary Architecture for STC UMS AS350B (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Title/Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 14.08.2013 |
| 10017243 | NSE INDUSTRIES S.A. | EASA.R.S.01440 | REV. 1 | | AS 365 N2, AS 365 N3, SA 365 N, SA 365 N1. | Installation of Brite Saver 2 Usage Monitoring System. Rev. 0. Revision of EASA.R.S.01440. Covers software upgrade and applicability modification. Rev. 1: Incorporates the definition file of the installation kit and different documentary changes | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: FAR Part 29, amends 29-0 through 29-55: Sec. 29.29, 29.301, 29.303, 29.30 (...) | NSE Industries AS365 Flight Manual Supplement doc. ref. RFMS-ARA-365 Rev. 02 dtd. 10 May 2013 or later revisions of the above listed documents approved by EASA. Conformity Statement and Documentary Architecture AS365, doc. ref. CONFAD-0079-92 Rev. 06 dtd. 30/04/2013. | None | 13.05.2013 |
| 10028220 | NSE INDUSTRIES S.A. | | REV. 2 | | 1- 206H, P206, P206A, P206B 1- P206C, P206D, P206E, T206H 1- TP206A, TP206B 1- TP206C, TP206D, TP206E 1- TU206A, TU206B, 1- TU206C, TU206D 1- TU206E, TU206G, TU206F 1- U206A, U206B, U206C, U206D 1- U206E, U206F, U206G 2- 206 | STC 10028220 revision 2 for STC DTS and DTS-V2 Cessna 206 - Installation Tracking System. Description: Change of Corporate Name and location, replacement of DTSV2 documentation reference, addition of TCDS USA A4CE to reflect all models listed, incorporated (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 23 Amtd 0-50. The requirements for environmental protection and the associated certified (...) | SDL-DTSV2-C206 Rev.00 dated 20/08/2013 AFMS-DTSV2-C206 Rev.00 or later revisions of the above listed documents approved by EASA | The approval holder shall fulfill the obligations of Part 21, Paragraph 21A.109. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or (...) | 05.08.2013 |
| 10029936 | NSE INDUSTRIES S.A. | | REV. 2 | | AS350B, AS350B1, AS350B2 AS350BA, AS350BB, AS350B3 | Installation of DTSV2 Data Transmission System. Revision 1 introduces corrections of the tracking system installation. Revision 2 is a re-print following a change of name and address. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Change complies with CS27 amtd 2. The requirements for environmental protection and the (...) | Aircraft Flight Manual Supplement DTSV2, doc. AFMS-Z4026-411683, Rev. 02 dated 27/04/2010; or later revisions of the above listed documents approved by EASA. STC Data List - "Supplement Type Certificate of DTS V2 installation on AS350" - doc. SDL-Z4018-411683 Rev. 07 dtd 26/06/2012. | Modification not approved for aircraft equipped with weather radar or AFCS (Automatic Flight Control System). Modification not approved for AS350B3 model helicopters with modification OP-4305 (Arriel 2D engine installation) incorporated. The approval ho (...) | 14.08.2013 |
| 10030143 | NSE INDUSTRIES S.A. | | | | AS355E, AS355F, AS355F1, AS355F2, AS355N, AS355NP | Installation of DTSV2 Tracking System | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | AFMS-Z4031-411683 revision 00 dated 12 March 2010 - Flight Manual Supplement or later revisions of the above listed documents approved by EASA. PCERT-Z4032-411683 revision 02 dated 3 May 2010 - Certification Plan; STDD-Z4027-411683 revision 00 dated 11 (...) | None | 27.05.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|--|--|--|--|---|---------------|
| 10030338 | NSE INDUSTRIES S.A. | P-EASA.R.S.01510 | | | AS350B, AS350B1, AS350B2, AS350BA, AS350BB, AS350B3 | Installation of a tracking system (DTS - Data Transmission System) | As per EASA TCDS.R.008 Issue 04, 23 November 2009The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval Change complies with CS27 Amdt 1The r (...) | Definition and installation-DTS Definition file, doc. DD-0018-053670-EN, rev. 02 dated 19/04/2010-DTS AS 350 and AS 355mounting Instruction, doc. IM-0011-053670-EN, rev. 08 dated 28/04/2010Inspection, maintenance, and operation-Operating and Maintena (...) | Modification not approved for aircraft equipped with weather radar orAFCS (Automatic Flight Control System) | 11.06.2010 |
| 10037841 | NSE INDUSTRIES S.A. | | REV. 2 | | EC135 P1, EC135 P2, EC135 P2+ EC135 T1, EC135 T2, EC135 T2+ EC635 P2+ EC635 T1 (CPDS), EC635 T2+ | Installation of a Brite Saver Usage Monitoring System. Thismodification is defined in CONFAD-0115-92-EN Rev01.Revision 1 - removes the limitation "only for EC135/EC635 variants withCPDS-Cockpit" | The Certification Basis for the original product remains applicable tothis certificate/ approvalThe requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval | Flight Manual Supplement RFMS-ARA-EC135-EN Rev. 00, dated22 December2011 or later revisions of the above listed documents approved by EASA- Conformity Statement and Documentary Architecture CONFAD-0115-92-ENRev. 01, dated 14 December 2012- Installa (...) | none | 14.01.2013 |
| 10042012 | NSE INDUSTRIES S.A. | | REV. 1 | | AS 350 B,AS 350 B1,AS 350 B2 AS 350 B3,AS 350 BA,AS 350 BB | Installation of Data Tracking System DTS-Alpha.Modification as defined in STC Data List, doc. SDL-DTSALPHA-AS350 Rev.04Revision 1 is a re-print following a change of name and address. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.This modification complies with CS-27 Amdt. 2 requirements as specifiedin Certificatio (...) | Definition and installation:Kit Part List, doc. NOM KIT-DTSA-03-00 Ind. 00Definition Document DTS Alpha P/N 412695 Amdt. B, doc. DD-412695 Rev.00STC Data List, doc. SDL-DTSALPHA-AS350 Rev. 04DTS Alpha Installation Manual on AS350, doc. IM-DTSALPHA- (...) | Modification not approved for rotorcraft equipped with AFCS(AutomaticFlight Control System) or Weather Radar.Modification not approved for AS350B3 model helicopters withmodification OP-4305 (Arriel 2D engine installation) incorporated.Prior to insta (...) | 14.08.2013 |
| 10043262 | NSE INDUSTRIES S.A. | | | | A109E | Installation of Night Image Vision System kit, allowing night VFR flight with Night Vision Goggles | The Certification Basis (CB) for the original product remains applicableto this certificate/ approval.The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | -ROTORCRAFT FLIGHT MANUAL SUPPLEMENT A109 , document RFMS-0020-12AF02Rev. 04or later revisions of the above listed documents approved by EASA- CONFORMITY STATEMENT AND DOCUMENTARY ARCHITECTURE A109, DocumentReference CONFAD-01A-1000-000 Rev. 01-INS (...) | Refer to applicable RFM for list of equipment mandatory forNight VisionImage System certificationInstallation of NVIS/NVG compatible lighting does not include or implyapproval for flight operations with Night Vision Goggles (NVG). Theoperator must r (...) | 17.01.2013 |
| 10038630 | NSE INTEGRATIONS | | | | ATR 42-300 | Modification in ATR42-300 MSN 049's calibration test bench | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalIn addition to the product's original certification basis, EASACertification Review Ite (...) | NSE Integrations's:Modification Form NSE 08 01 Edition 3, Revision 1 dated 20th September2011 andSupplement for EWIS Instructions for Continued Airworthiness: "DetailedInspection Tasks ATR42 Kit Calibration Test Bench" revision Indice B,dated 20th Se (...) | This STC is limited to ATR42-300 serial number aircraft MSN049 | 28.09.2011 |
| 10015792 | O&N AIRCRAFT MODIFICATIONS, INC. | EASA,IM.A.S.01851 | | FAA STC SA615NE | CESSNA 172F CESSNA 172G CESSNA 172H CESSNA 172I CESSNA 172L CESSNA 172M CESSNA 172N CESSNA 172P | Installation of an 18 gallon fuel tank in the aft baggage compartment, and fuel transfer system. | - Installation in accordance with O & N Installation Drawing List 172100, Report No ON-1012, Revision A, dated June 1 1989, or later approved revision.- Eligible on aircraft serial numbers as identified on the Approved Flight Manual Sup (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 29.08.2006 | |
| 10015986 | O&N AIRCRAFT MODIFICATIONS, INC. | EASA,IM.A.S.02139 | | FAA STC SA1035NE | CESSNA 150F CESSNA 150G CESSNA 150H CESSNA 150J CESSNA 150K CESSNA 150L CESSNA 150M CESSNA 152 | Installation of a 14.6 gallon aft fuselage tank in baggage compartment. | Installation in accordance with drawing list No. ON-1025 rev. B, dated 21-07-92, or later approved revisions.AFM-Supplements as mentioned below | 1. FAA Approved Supplement to Pilots' Operating Handbook and/ or FAA Approved Airplane Flight Manual, Nr. 19, 20 and 21, dated Jan. 13-93, or later approved revision, is requiredCessna 150F, G, H, J, K, L & M and 152.2. This installation is not elig (...) | 13.10.2006 | |
| 10043070 | O&N AIRCRAFT MODIFICATIONS, INC. | EASA,IM.A.S.03167 | | FAA STC SA02277CE | M20F, M20J | Installation of Fuel Bladder Cells in wet wing fuel cavities. | Installation and maintenance i.a.w. to- Configuration 1: Brentwing Engineering, Inc. Installation Drawing ListB-400 and/ or Installation Instructions Report B-200 and MaintenanceInstructions Report B-300, all in latest approved Revisions- Configur (...) | none | 07.05.2009 | |
| 10043071 | O&N AIRCRAFT MODIFICATIONS, INC. | EASA,IM.A.S.03166 | | FAA STC SA02277CE | M20C, M20D, M20E, M20G | Installation of Fuel Bladder Cells in wet wing fuel cavities. | Installation and maintenance i.a.w. to- Configuration 1: Brentwing Engineering, Inc. Installation Drawing ListB-700 and/ or O->(&-<)-N Aircraft Modifications, Inc. InstallationInstructions B-900 and O->(&-<)-N Aircraft Modifications, Inc.Maintenance I (...) | none | 07.05.2009 | |
| 10015797 | OFEK HOLDING CORPORATION | EASA,IM.A.S.01858 | | FAA STC SA01415CH | PIPER PA-31T | Installation of camera provisions in fuselage floor cut-out | CAA-NL report Nr. LTB/TCP/06.531118 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 30.01.2006 | |
| 10015798 | OFEK HOLDING CORPORATION | EASA,IM.A.S.01859 | | FAA STC SA01961CH | PIPER PA-31T | Installation OFEK Holding Company camera provisions installed pursuant to STC SA01962CH | CAA-NL report Nr. LTB/TCP/06.531118 | | 30.01.2006 | |
| 10034682 | OGMA, INDUSTRIA AERONAUTICA DE | | | | EMB-135BJ | Change Number 1086.07 - "Installation of coffee maker & associated watertank" | The Certification Basis for the original product remains applicable tothis certificate/ approval.The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | - Change approval sheet: CAS-A1.50C-1086.07-25.00-001 dated14Apr2011- Change Dossier: CD-A1.50C-1086.07-25.00-001 Rev.A dated 14Apr2011- Aircraft Maintenance Manual Supplement: MS-A1.50C-1086.07-00.00-002initial issue, dated 23Nov2010,or later revis (...) | none | 19.04.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|----------------------|--|--|--|--|--|---------------|
| 10037579 | OGMA, INDUSTRIA AERONAUTICA DE | | | | EMB-145EP, EMB-145LR, EMB-145LU | Installation of a VIP meeting room on EMB-145 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Change approval sheet doc. N° 10 P15 CAS 02 001, Rev. Orig. dated 24.11.2011 - Change Dossier doc. N° 10 P15 CD 02 001, Rev. Orig. dated 21.11.2011 - Airplane Flight Manual Supplement doc. N° 10 P15 MS 02 010, Rev. Orig. dated 18.11.2011 - Maintenance M (...) | Aircraft S/N 145076, 145084, 145129, 145135, and 145147. | 08.12.2011 |
| 10039843 | OGMA, INDUSTRIA AERONAUTICA DE | | | | EMB-145EP, EMB-145EU, EMB-145LR, EMB-145LU | Nespresso Coffe Maker Installation Description as per Change Dossier, ref. 11P05CD00001 initial issue <-> dated 23th May 2012. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Change Dossier, ref. 11P05CD00001 initial issue, dated 23th May 2012; or later revisions of the above listed documents approved by EASA. | Applicability for the following S/N: 145076, 145084, 145129, 145135, 145147 <-> 145242, 145258. | 29.05.2012 |
| 10042497 | OGMA, INDUSTRIA AERONAUTICA DE | | | | EMB-145EP, EMB-145LR, EMB-145LU | VIP room extension and Iridium Satphone system installation on EMB-145 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Change Dossier doc. N° 12P06 CD 00 001, Rev. Orig. dated 30.11.2012 - Airplane Flight Manual Supplement doc. N° 12P06 MS 00 010, Rev. Orig. dated 15.11.2012 - Maintenance Manual Supplement doc. N° 12P06 MS 00 020, Rev. Orig. dated 30.11.2012 or later (...) | The change is applicable to the following aircraft having OGMA STC10037579 incorporated, and that do not have flux valve sensors: A/C SN145076, 145084, 145129, 145135, and 145147 | 05.12.2012 |
| 10031182 | OHS AVIATION SERVICES GmbH | EASA.A.S.03940 | | | 328-100, 328-300 | Change seat cushions bottom and backrest, JB6 series and AMP1300 series for DO 328-100 and -300 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | OHS Master Documents List ASE04-10-MDL-01-xx, Revision 4 dated 18.06.2010. Applicable revisions or later revisions of the above listed documents approved by EASA | None | 30.07.2010 |
| 10015480 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.A.S.01238 | | FAA STC ST10395SC | MD-11 | EASA Validation of FAA STC ST10395SC on #Installation of two non-taxi, take-off and landing (TT7L) courier seats* and associated Airplane Flight Manual Supplement | | FAA STC ST10395SC dated 1 April 2005. Aerote Master Document List Ref MDL-407, Revision C, 31 March 2005. Airplane Flight Manual Supplement Ref FMS-411, Revision IR, 1 April 2005 | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 31.05.2005 |
| 10016706 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.00457 | REV. 1 | FAA STC NO SR01424SE | AS355 E, AS355 F, AS355 F1, AS355 F2, AS355 N, AS355 NP | Cargo Hook Kit model 200-292-01 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 155-104-00 revision 11 dated 9 April 2010 - Master Drawing List: 120-112-01 revision 4 dated 28 May 2010 - Owners Manual: 123-017-01 revision 4 dated 28 May 2010 - Instructions for Continued Airworthiness: 122-005-00 revision 17 dated 9 March 2010 - Car (...) | None | 21.07.2010 |
| 10016709 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.00553 | REV. 1 | FAA STC SR00699SE | BELL 204B, BELL 205A, BELL 205A-1, BELL 212, BELL 412, BELL 412 EP | To amend EASA STC EASA.IM.R.S.00553 to list the correct document number for the Rotorcraft Flight Manual Supplement (RFMS) Ass. Section 4 of Onboard Systems Service Manual No. 122-004-00 Rotorcraft Flight Manual Supplement No. 121-020-00 dated 24 September 2003 or later approved revision | | - FAA STC SR00699SE-RFM No 120-083-00 Section 4 of Onboard Systems Service Manual No. 122-004-00 Rotorcraft Flight Manual Supplement No. 121-020-00 dated 24 September 2003 or later approved revision | Approval of this change in type design applies only to Bell 204B, 205A, 205A-1, 212, 412 and 412EP rotorcraft which where previously equipped with an approved installation of the following cargo hook: Bell P/N 204-072-924-1 or Breeze Eastern P/N SP7109-1, -11 or-62. | 19.10.2005 |
| 10016711 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.00623 | | FAA STC SR01166SE | EUROCOPTER FRANCE AS 350 B3 | Cargo Hook Kit installation, in accordance with FAA STC No. SR1166SE issued January 22, 2003. | | - AS 350 B3 Flight Manual and RFM Supplement Onboard No. 121-014-00 dated January 15, 2003, and revised on July 23, 2004 or later approved revisions. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 15.03.2005 |
| 10016726 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01027 | REV. 2 | FAA STC SR00886SE | AS350B, AS350BA, AS350B1, AS350B2, AS350B3, AS350D, EC130B4 | Installation of Cargo Hook Kit Model 200-261-00, 200-261-01, and 200-353-00. This STC is a revision to EASA STC EASA.IM.R.S.01027 Revision 1 through the validation of amendment dated 11 August, 2009 to FAA STC SR00886SE. | The original product certification basis as per EASA TCDS R.008 Issue 04, 23 November 2009. The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Definition and installation: Onboard Systems Master Drawing List Doc. No. 155-060-00, Rev. 11, dated June 11, 2009. Installation of the 200-261-00 Cargo Hook Kit in accordance with Onboard Systems Owner's Manual No. 120-094-00, Rev. 7 dated September 05, (...) | The cargo hook kits approved under this STC can be installed on AS 350B, BA, B1, B2, and D helicopters equipped with Eurocopter swing or sling type systems, or on AS350B3 equipped with Eurocopter sling type system only, or on EC 130B4 equipped with Eu (...) | 13.01.2010 |
| 10016781 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01122 | | FAA STC SH1262NW | EUROCOPTER FRANCE AS 350 B, EUROCOPTER FRANCE AS 350 B1, EUROCOPTER FRANCE AS 350 B2, EUROCOPTER FRANCE AS 350 BA, EUROCOPTER FRANCE AS 350 D, EUROCOPTER FRANCE AS 350 D1 | Electronic Model E-69 Cargo Hook Load Cell System p/n 200-295-00, in accordance with Onboard Systems Master Drawing List n° 155-010-00, revision 10 dated 15/06/04 approved by FAA, or later FAA approved revision. | | - Installation Instructions : Onboard Systems Owners Manual n° 120-114-00, revision 0 dated 15/06/04 approved by FAA, or later FAA approved revision - Instructions for Continued Airworthiness : section 4 of Onboard Systems Owners Manual n° 120- (...) | 1 - The present STC is the validation of the FAA STC SH1262NW last amended on 15/03/05, limited to Eurocopter AS 350 B, BA, B1, B2, and D models. 2# The system covered by this STC can be installed on AS 350 helicopters equipped with Eurocopter sling s (...) | 10.01.2007 |
| 10016789 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01136 | | FAA STC SH4908NM | MD HELICOPTERS 369D, MD HELICOPTERS 369E | Installation of Model E-51 Cargo Hook Load Cell System PN 200-034-00. This STC is the validation of the FAA STC SH4908NM amended on 05/11/1990 and limited to MDHI 369 D and E models | | Master Drawing List (MDL) Onboard Systems MDL No. 155-014-00 Revision 1 dated 10/10/1990 approved by FAA or later FAA approved revision. Installation Instructions Onboard Systems Owners Manual No. 120-017-00 Revision 1 dated 10/10/1990 approved by FAA (...) | 1. This STC is to be installed on aircraft models specified in "applicability" section above, equipped with MDHI Cargo Hook Kit No. 369H90072, using MDHI cargo hook manual release PN 14323-2 only. 2. Installation, Inspection and Operation must be done (...) | 12.10.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|----------------------|--|---|---|--|---|---------------|
| 10016804 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01164 | | FAA STC SR01588SE | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 D | ONBOARD SYSTEMS fuel drain guard kit for Eurocopter AS 350 series helicopter.Validation of FAA STC SR01588SE | | Refer to installation instructions, inspection and maintenance instructions listed in FAA STC SR01588SE reissued September 29, 2005 | Refer to FAA STC SR01588SE reissued September 29, 2005 | 06.03.2006 |
| 10016806 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01166 | | FAA STC SR00369SE | BELL 206L BELL 206L-1 BELL 206L-3 BELL 206L-4 BELL 407 | Cargo hook installation n. 200-215-00 | | - FAA approved Onboard Systems Master Drawing List n. 155-029-00 dated June 26, 2000 or later FAA approved revision, -FAA approved Onboard Systems Owner's manual n. 120-046-00 dated June 26, 2000, or later FAA approved revision, - FAA approved Onbo (...) | Approval of this change in type design applies to only those Bell model rotorcraft listed above, which were previously equipped with an FAA approved installation of Bell cargo hook suspension systems 206-706-341-5, 206-706-341-101; 206-706-341-109 a (...) | 27.09.2006 |
| 10016807 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01167 | | FAA STC SR00892SE | MD HELICOPTERS 369D MD HELICOPTERS 369E MD HELICOPTERS 369F MD HELICOPTERS 369FF MD HELICOPTERS 369HE MD HELICOPTERS 369HM MD HELICOPTERS 369HS MD HELICOPTERS 500N | Installation of Onboard Systems Model 200-264-00 Cargo Hookkit.EASA validation of FAA STC n° SR00892SE, amended January 13th, 2003, or later approved revision.This STC is classified 'Basic' by the FAA and is approved under the post type Validation Principles (PTVP). | | Onboard Systems Master Drawing List n°155-061-00, revision 7 dated June 18th, 2008 or later approved revision Onboard Systems Owner's Manual n°120-096-00, revisiona dated September 22nd, 2005 or later approved revision.Onboard Systems Instruction fo (...) | This cargo hook is not approved for human external load.Approval of this installation applies to the above listed model rotorcraft only, which were previously equipped with an approved installation of McDonnell Douglas cargo hook suspension systems a (...) | 08.07.2008 |
| 10016809 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01169 | | FAA STC SR00557SE | ROBINSON R22 ROBINSON R22 ALPHA ROBINSON R22 BETA ROBINSON R22 MARINER | Cargo Hook Suspension System | | FAA approved Onboard Rotorcraft Flight Manual Supplement No121-022-00, dated September 29, 2005 or later approved revisionFAA approved Onboard Systems Owner's Manual No 120-070-00, dated October 11, 2004 or later approved revision | | 02.01.2006 |
| 10016811 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01171 | REV. 1 | FAA STC SR01064SE | ROBINSON R44 ROBINSON R44 II | Cargo hook suspension system i.a.w. US STC SR001064SE.Note:Revision 1 of this STC is made for correction of FAA original STC reference. This supersedes previous issuance of the STC. | | - FAA approved Onboard Systems Rotorcraft Flight Manual Supplement No. 121-007-00, dated April 22, 2005 or later approved revision for the Robinson R44 rotorcraft- FAA approved Onboard Systems Rotorcraft Flight Manual Supplement No. 121-017-00 (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 29.04.2009 |
| 10016862 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01238 | | FAA STC SR01164SE | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 D | Fabrication and Installation of Onboard Systems Model 200-280-01 Cargo Hook Swing Suspension System | | - FAA STC SR01164SE dated October 26, 2004,- Master Drawing List STC SR01164SE Doc. Number 155-086-00 dated September 29, 2004 Revision 10,- Owner's Manual for the Cargo-Hook Swing Suspensions system on the Eurocopter AS350 Series System, P (...) | 1. Approval of this change in Type Design applies only for those Eurocopter AS350 model rotorcraft listed above;2.This approval should not be extended to other rotorcraft of these models on which other previously approved modifications are incorporated unless it is determined by the installer t | 26.07.2006 |
| 10016869 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01245 | | FAA STC SR00649SE | BELL 206A BELL 206B | Cargo Hook Suspension Kit for Bell 206A and 206B | | 200-233-00 or 200-235-00 (without load weigh).200-234-00 or200-236-00 (with load weigh).MDL 155-041-00 rev 1 dated November 20, 1998 or later approved revision Owners Manual 120-076-00 revision 1 dated November 20, 1998 or later approvedrevision Flight Manual Supplement 120-076-00 dated December | None | 21.12.2006 |
| 10016870 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01246 | REV. 1 | FAA STC SR01393SE | AS350B, AS350BA, AS350B1 AS350B2, AS350B3, AS350D | Onboard Systems cargo hook kits P/N 200-286-01 (with cargo hook P/N528-023-01) and P/N 200-286-02 (with cargo hook P/N 528-029-00)Revision 1 - introduces cargo hook kit P/N 200-286-02This STC is the validation of FAA STC SR01393SE (issued on April 05,2004 and last amended on June 29, 2010). | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approval.Change under revision 1 complies with FAR Part 27, amds 27-1 through27-39The requirem (...) | Definition and installation:Onboard Systems Master Drawing List No. 155-093-00, Rev. 8 datedNovember 16, 2010 and/or cargo hook kit P/N 200-286-01;Onboard Systems Owner's Manual No. 120-107-01 Rev. 6 dated September 01,2010for cargo hook kit P/N 200 (...) | None | 04.02.2011 |
| 10016874 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01250 | | FAA STC SR00920SE | ROBINSON R22 ROBINSON R22 ALPHA ROBINSON R22 BETA ROBINSON R22 MARINER | Cargo Hook Suspension Kit i.a. with FAA STC SR00920SE | | Rotorcraft Flight Manual Supplement No. 121-004-00, Rev. 1,FAA approved: June 9, 2006 or later FAA approved revisionOwner's Manual No. 120-095-00, Rev. 0, dated January 12, 2001, or later approved revisionService Manual No. 122-012-00, Rev. 1, dated (...) | | 20.12.2006 |
| 10016907 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01287 | | FAA STC SR00711SE | SCHWEIZER 269 SERIES | Validation of FAA STCSR00711SE , amended January 13, 2003, or later approved versions. | | Fabrication of Onboard Systems Model 200-244-00 cargo hook kit i.a.w. FAA approved onboard system MDL No 155-045-00, dated 3 May 1999, or later approved revision.Installation of this replacement cargo hook i.a.w. FAA approved Onboard Systemowner#s ma (...) | Approval of this STC apply only to those Schweizer 269C which were previously equipped with FAA approved installation of Schweizer cargo hook kit P/N 269A4971-27.Modified rotorcraft must be operated i.a.w. FAA approved Rotorcraft Flight manual Suppl (...) | 09.07.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|----------------------|--|---|---|---|---|---------------|
| 10016937 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01324 | REV. 2 | FAA STC SR01164SE | AS 350 B, AS 350 B1 AS 350 B2, AS 350 B3 AS 350 BA, AS 350 D | External Load Swing Suspension System models 200-280-01 and-04 (with Keeperless Cargo Hooks) and model 200-280-02 and -03 (with Hydraulic Cargo Hooks) Revision 1 - introduced cargo hook kit 200-280-03 with a collectiveriction knob and a new release lev (...) | As per EASA TCDS.R.008 Issue 04, 23 November 2009. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Change under Revision 2 complies wit (...) | Definition and installation: Onboard Systems Master Drawing List No. 155-086-00, Rev. 27, dated October 30, 2009, approved by FAA. Onboard Systems Owner's Manual No. 120-104-01 (for system model 20028001). Rev. 10 dated October 30, 2008, approved by FAA (...) | 1. EASA-approved AS350 Flight Manual and-External Load Transport: Cargo Swing-FM supplement are required. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Com (...) | 21.06.2010 |
| 10016990 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01396 | REV. 1 | FAA STC SR0407SE | MD HELICOPTERS 369D MD HELICOPTERS 369E MD HELICOPTERS 369F MD HELICOPTERS 369FF MD HELICOPTERS 369HE MD HELICOPTERS 369HM MD HELICOPTERS 369HS MD HELICOPTERS 500N | Onboard Systems Model 200-187-00 cargo hook on helicopters 369D, 369E, 369FF, 369HE, 369HM, 369HS, 500NEASA validation of FAA STC SR0407SE, amended 2 October 2007, or later approved 07This STC is classified #Basic by the FAA and is approved under the post type Validation Principles (PTVP) | | Fabrication of kit in accordance with FAA Approved Onboard Systems Master Drawing List No. 155-030-00, Revision 12, dated February 22, 2008, or later FAA approved revision. Installation of this replacement cargo hook in accordance with FAA approved O (...) | Approval of this change in type design applies to only those McDonnell Douglas model rotorcraft listed above, which were previously equipped with an FAA approved installation of McDonnell Douglas cargo hook suspension systems and Breeze-Eastern cargo (...) | 18.03.2008 |
| 10016993 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01402 | | FAA STC SR01795SE | EUROCOPTER FRANCE EC 120 B | Installation of Cargo Hook Suspension Kit P/N 200 - 308 # 00. | | 1. Definition and Installation in accordance with:- Onboard Systems Model 200-308-00, Cargo Hook Suspension Kit;- Onboard Systems Master Drawing List 155-115-00, Revision 1, dated October 22, 2007 or later approved Revision;- Onboard Systems (...) | 1. All limitations of RFM supplement apply. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse (...) | 22.01.2008 |
| 10016999 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01409 | | FAA STC SR01778SE | MD HELICOPTERS 369D MD HELICOPTERS 369E MD HELICOPTERS 369F MD HELICOPTERS 369FF MD HELICOPTERS 369HE MD HELICOPTERS 369HM MD HELICOPTERS 369HS MD HELICOPTERS 500N | Onboard Systems Model 200-300-00 talon LC Hydraulic Cargo Hook Kit without load weight, Model 200-301-00 talon LC Hydraulic Cargo Hook Kit with load weight on helicopters 369D, 369E, 369FF, 369HE, 369HM, 369HS, 500N | | Onboard Systems Master Drawing List No. 155-112-00, Revision 2, dated 4 March 2008, or later approved revision; Onboard Systems Owner's Manual (for 200-300-00 Cargo Hook Kit) No. 120-119-00, Revision 1, dated 13 July 2007, or later approved revisions. (...) | This cargo hook is not approved for human external load. Approval of this change in type design applies to only those McDonnell Douglas model rotorcraft listed above, which were previously equipped with # Onboard system model 200-187-00 cargo hook (a (...) | 08.07.2008 |
| 10017020 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01438 | REV. 1 | FAA STC SR01808SE | R44, R44 II | Cargo Hook System Kits The installation as defined in MDL 155-124-00 provides for different optional Cargo Hook System kits depending on aircraft voltage system, with and without load weighing system and the respective cargo hook in accordance with FAA STC SR01808SE | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Flight Manual Supplements:- For the 200-324-00 and 200-325-00 cargo hook kits: Rotorcraft Flight Manual Supplement (RFMS) No. 121-051-00, Revision 1, dated February 25, 2009. For the 200-326-00, 200-327-00 and 200-327-01 cargo hook kits: RFMS No. 121-048- (...) | Approval of this change in type design applies only to those Robinson rotorcraft models listed above which are equipped with Robinson hardpoint/ tie down block P/N 134-1. | 17.12.2009 |
| 10017025 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01444 | | FAA STC SR01815SE | EUROCOPTER FRANCE EC 130 B4 | Installation of Model 200-321-00 Cargo Hook Swing Suspension Kit and Model 200-322-00 Cargo Hook Sling Suspension Kit. | | # Definition and installation: # Definition and installation: # Onboard Systems Master Drawing List No. 155-121-00, Rev. 1 dated February 6, 2008 or later EASA approved revisions; # Onboard Systems Owner's Manual No. 120-131-00 Rev. 1 dated (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 27.03.2009 |
| 10017027 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01446 | | FAA STC SR01812SE | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 D | Installation of Cargo Hook Kit P/N 200-297-00 and installation of Cargo Hook Kit P/N 200-298-00 This STC is the validation of FAA STC SR01812SE | | Definition and installation: Onboard Systems Master Drawing List Doc. No. 155-110-00, Rev. 1, dated September 27, 2006, or later EASA approved revisions; Onboard Systems Owner's Manual No. 120-115-00, Rev. 0 dated April 25, 2005, or later EASA approved (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 21.11.2008 |
| 10017037 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01461 | | FAA STC SR01394SE | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 D | Installation of Cargo Hook Sling Suspension Kit P/N 200-287-00 This STC is the validation of FAA STC SR01394SE | | Definition and installation: # Onboard Systems Master Drawing List Doc. No. 155-096-00, Rev. 5, dated March 4, 2008, or later EASA approved revisions; # Onboard Systems Owner's Manual for the Cargo Hook Sling Suspension System Retrofit Kit for th (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 06.11.2008 |
| 10017080 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.01531 | REV. 1 | FAA STC SR01943SE | BELL 407 | 200-328-00 Cargo Hook Fixed Provisions Kit; 200-329-00 Cargo Hook Suspension Kit with Pillow Blocks; 200-330-00 Cargo Hook Suspension Kit without Pillow Blocks; 200-331-00 Pin Load Weigh Kit | | 155-125-00 revision 1 dated 16 December 2008, or later EASA approved revision # Master Drawing List: 120-136-00 revision 1 dated 15 December 2008, or later EASA approved revision # Owners Manual; 123-032-00 revision 1 dated 15 December 2008 or later EASA approved revision # Instructions for Continued | 1. For P/N 200-329-00 (Cargo Hook Suspension Kit with Pillow Blocks) # P/N 200-328-00 (Fixed Provisions Kit) or Bell Helicopter P/N 206-706-341-111, -117 or -123 (Auxiliary Equipment Kit # Cargo Hook Provisions) must be installed; For P/N 200-330-0 (...) | 10.03.2009 |
| 10028873 | ONBOARD SYSTEMS INTERNATIONAL | | | FAA STC SR02028SE | EC 120 B | Cargo hook kit Model 200-351-00 | As specified in TCDS DGAC No 189 The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Definition and Installation:- Onboard Systems Master Drawing List No. 155-144-00, Rev. 0, dated July 22, 2009- Onboard Systems Owner's Manual No. 120-139-00, Rev. 0, dated June 16, 2009 Inspection and Maintenance:- Onboard Systems Instructions for Cont (...) | EASA-approved EC 120 B Flight Manual External Load Transport "Cargo Sling" FM Supplement is required. | 16.02.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|---------------------|----------------|----------------------|---|--|---|--|---|---------------|
| 10030504 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.00542 | REV. 1 | FAA STC SR01165SE | AS 350 B, AS 350 B1, AS 350 B2 AS 350 B3, AS 350 BA, AS 350 D | Revision of STC EASA.IM.R.S.00542 introducing additional cargo hook kit200-282-03. | As specified in TCDS R.008The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Definition and Installation:- Onboard Systems Master Drawing List No. 155-087-00, Rev. 13, dated October 30, 2009.For kit P/N 200-282-03, Onboard Systems Owners Manual No. 120-105-00, Rev. 12, dated June 20, 2008.For kit P/N 200-282-03- Onboard System (...) | EASA-approved AS 350 Flight Manual and-External Load Transport: Cargo Slings-FM supplement are required.This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description" Compatib (...) | 21.06.2010 |
| 10030532 | ONBOARD SYSTEMS INTERNATIONAL | P-EASA.IM.R.S.00593 | REV. 1 | FAA STC SR000896SE | BELL 206 A, 206 B | Cargo Hook Kit | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 155-063-00 revision 7 dated 23 October 2009 - Master Drawing List;120-098-00 revision 6 dated 4 August 2009 - Owners Manual for200-267-00 Cargo Hook Kit;-122-005-00 revision 9 dated 3 January 2006 - Service manual for200-267-00 Cargo Hook Kit;-120 (...) | None | 23.06.2010 |
| 10030772 | ONBOARD SYSTEMS INTERNATIONAL | | | FAA STC SR20035SE | AS355E, AS355F, AS355F1, AS355F2, AS355N, AS355NP | Cargo Hook Kit model 200-352-00 | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 155-145-00 revision 1 dated 28 May 2010 - Master Drawing List;120-140-00 revision 0 dated 9 September 2009 - Owners Manual;123-035-00 revision 0 dated 18 November 2009 - Service Manual;121-054-00 revision 1 dated 11 June 2010 - Rotorcraft Flight Manual Supplement. | None | 07.07.2010 |
| 10032737 | ONBOARD SYSTEMS INTERNATIONAL | EASA.IM.R.S.00592 | REV. 1 | FAA STC SR00850SE | 206L, 206L-1, 206L-3, 206L-4 407 | Cargo Hook Kit model 200-260-00 and 200-260-01This change adds an additional cargo hook | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 155-057-00 revision 19 dated 30 October 2009 - Master Drawing List;120-093-00 revision 8 dated 15 September 2009 - Owners Manual;123-001-00 revision 4 dated 11 August 2009 - Instructions for Continued Airworthiness;122-005-00 revision 15 dated 23 Apr (...) | Applies only to those Bell Helicopter models listed above previously equipped with Bell Helicopter Cargo Hook Suspension System206-706-341-5, 206-706-341-101 or 206-706-341-109 and either Breeze-Eastern cargo hooks 17149-2, 17149-6 or Onboard Systems Cargo hooks 528-010-04 or 528-023-01 | 26.11.2010 |
| 10040250 | ONBOARD SYSTEMS INTERNATIONAL | | | FAA STC SR02217SE | EC135 P1(CDS), EC135 P1(CPDS) EC135 P2 (CPDS), EC135 P2+ EC135 T1(CDS), EC135 T1(CPDS) EC135 T2(CPDS), EC135 T2+ EC635 T1 (CPDS), EC635 P2+ EC635 T2+ | Cargo Hook Kit. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement 121-057-00, Revision 0, FAA Approved 27 February 2012; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)- Mast (...) | Limited to helicopters equipped with cargo hook assembly P/N AS22-44. | 20.06.2012 |
| 10040440 | ONBOARD SYSTEMS INTERNATIONAL | | | FAA STC SR01694SE | BO 105 S, BO 105 LS A-3 | Validation of FAA STC SR01694SE - Cargo Hook Kits | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.FAR 27 including amendment 37The requirements for environmental protection and the asso (...) | FAA approved Onboard Systems Rotorcraft Flight Manual Supplement RFMS121-029-00, dated March 1, 2012; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this dec (...) | This STC is restricted to installation of cargo hook kits model200-387-00 and model 200-388-00. Not included are cargo hook kits model200-302-00 and model 200-303-00. | 03.07.2012 |
| 10043044 | ONBOARD SYSTEMS INTERNATIONAL | P-EASA.IM.R.S.01530 | | FAA STC SR00525SE | MD HELICOPTERS 369D, 369E 369F, 369FF, 369HE, 369HM 369HS, 500N, 600N | Installation of Onboard Systems Model 200-231-00 and 200-231-01 Cargo Hook Attach Point Kits in accordance with Onboard Systems Master Drawing List No. 155-038-00, Rev. 6, dated August 26, 2008, or later approved revision. | | - Installation: Installation of the Cargo Hook Attach Point Kits must be done in accordance with FAA-approved On board Systems Owner's Manual No.120-073-00, Rev. 4, dated August 25, 2008, or later approved revision.- Inspection: Inspection of the Atta (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.The inst (...) | 08.06.2009 |
| 10038364 | ONE SKY AVIATION | | | FAA STC SA02372AK | 1. B200, B200C, B200CT, B200T 1. B200GT, B200CGT 1. B300, B300C 2. C90GT, C90GTI | Installation of the Max-Viz® Enhanced Vision System (EVS) Model-1500 on Hawker Beechcraft Corporation B200/B300/C90 Series aircraft equipped with a Rockwell Collins ProLine 21 Adaptive Flight Display (AFD) having the video capability upgrade Part No 82 (...) | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Installation in accordance with One Sky Aviation, LLC Master Data Listref Doc No 20 11 00, Rev 3, dated July 19, 2010, Operation in accordance with Aircraft Flight Manual Supplement, One Sky Aviation, LLC document number 32 11 04, Rev B, dated July 1 (...) | None | 09.02.2012 |
| 10040870 | ORCON CORPORATION | | | | B737-200/-200C/-300/-400/-500 | Thermal/acoustic insulation system for Boeing 737 series aircraft.insulation blankets installation. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.CS25.856(a) (amendment 6) Thermal/acoustic insulation materialsThe certificated noise (...) | Master Drawings List MDL-8000 revision H<(>,<)->Continued Airworthiness instructions CAI-8002 revision E<(>,<)->or later revisions of the above listed documents approved by EASA | None | 31.07.2012 |
| 10041532 | ORCON CORPORATION | | | FAA STC ST01834LA | B757-200/-200PF | Validation of FAA STC ST01834LA related to Installation of Insulation Blanket | The Certification Basis for the original product remains applicable to this certificate/ approval. | Orcon Corporation's Index List, Document No. F83005, Revision H, dated 23 April, 2012; Orcon Corporation's Master Drawing List, Document No. MDL-9000, Revision H, dated April 23, 2012; Orcon Corporation's Continued Airworthiness Instructions, Document No. (...) | None | 25.09.2012 |
| 10042039 | ORCON CORPORATION | | | FAA STC ST02350LA | CL600-2B19 | Orcon Thermal/Acoustic Insulation System for Bombardier Challenger 850 Series Aircraft | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. ST02350LA dated 6th May 2011, Orcon Corporation's Index List, No. F83008, Revision B, dated April 20, 2011, Orcon Corporation's Continued Airworthiness Instructions, Document No. CAI-CL850 Revision A, dated January 5, 2011 or later revisions o (...) | None | 05.11.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|----------------------|---------------------------------------|---|--|---|--|---------------|
| 10027300 | OREGON AERO, INC. | | | FAA STC SA01597SE | LC41-550FG (C400) | Installation of replacement seats | CS-23The Certification Basis for the original product remains applicable to this certificate/ approvalJAR 23, Amdt. 1 as referred to EASA TCDS IM.A.516The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | -According to Master Drawing List 6MDL-HG 10, Rev.J.04 and-operated acc. to AFM-S Oregon Aero High-G Seats SAFM-C350/400-0001-maintained law. Engineering Manual Instruction for ContAW EM-01-01, Rev.B or later revisions of the above listed documents ap (...) | as indicated in Engineering Manual Instruction for ContAW EM-01-01, RevB, or later FAA approved revisions-only for LC41-550FG SN 41002 through 41 108, 41501 through 41800 and411001 and onwards | 18.09.2009 |
| 10016451 | PANASONIC AVIONICS CORPORATION | EASA.IM.A.S.02797 | | - | BOEING 757-200 SERIES | Installation of a Matsuhita Avionics Audio System on a Boeing 757-200 Aircraft. | | E.J. Mlynarczyk Drawing List DL10053, Revision H, 26/09/1994 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 18.07.2008 |
| 10016826 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01196 | | FAA STC SR00338DE | BELL 407 | Installation of Anti-Torque Extension Adjustment System | | Master Drawing List DL-407PX-100 rev B | None | 30.06.2006 |
| 10016875 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01252 | | FAA STC SR00375DE | BELL 206A/B/L/L-1/L-3/L-4 BELL 206L-4 | Laser Mapping Installation | | Master Drawing List DL-206LM-100 rev N/C dated March 5, 1999, or later approved revisionFlight Manual Supplement PR-206LM-110B dated April 12, 1999, or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 15.09.2006 |
| 10016949 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01344 | | FAA STC SR00489SE | BELL 407 | Installation of Cabin Air Conditioner System. | | DL-407AC-110 revision A dated 24 November 1997 or later approved Revision # Master Drawing List.PR-407AC-110M revision 3 dated 24 July 2003 or later approved revision # RotorcraftFlight Manual Supplement.PR-407AC-120M revision 7 dated 1 August 2006 (...) | | 21.08.2007 |
| 10016950 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01345 | | FAA STC SR00280DE | BELL 407 | Installation of Bleed Air Cabin Heater System. | | DL-407H-100 revision D dated 24 February 1997 or later approved Revision # Master Drawing List.PR-407H-100M revision 2 dated 18 April 2002 or later approved revision # Rotorcraft Flight Manual Supplement.PR0407H-120M revision N/C dated 14 April 1997 (...) | | 21.08.2007 |
| 10016952 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01348 | | FAA STC SR00025DE | BELL 206A/B/L/L-1/L-3/L-4, 407 | Installation of an Equipment Tray. | | DL-206/407ET-100 revision N/C dated 27 July 1999 or later approved Revision # Master Drawing List. | This approval is for a shelf only. The shelf is rated for a15 pound load and must be placarded as shown in drawing 206ET-1000, Revision B dated 9 February 1993 or later approved revision.On 206A and B models, this approval is for serial numbers 154 and subsequent due to airframe differences. | 21.08.2007 |
| 10016953 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01349 | | FAA STC SR00335DE | BELL 407 | Snow Pads Installation. | | - DL-407SP-100 revision A dated 22 January 1998 or later approved Revision # Master Drawing List - PR-407SP-100M dated 2 February 1998 or later approved revision # Rotorcraft Flight Manual Supplement | | 21.08.2007 |
| 10016954 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01350 | | FAA STC SR00344DE | BELL 407 | Installation of External Mirror. (This STC has been classified as #BASIC# by the FAA in accordance with FAA Order 8110.52 and has been approved using the PTVP Procedures.) | | DL-407EM-100 revision N/C A dated 9 January 1998 or later approved Revision # Master Drawing List. | | 06.09.2007 |
| 10016955 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01351 | | FAA STC SR00011DE | BELL 206B/L/L-1/L-3/L-4, 407 | Provision for Mount of a Video Monitor Placed on Left or Right Instrument Panel Console. | | DL-206/407MM-110 revision N/C dated 27 July 1999 or later approved Revision # Master Drawing List. | This approval is for the Mount Assembly only. The monitor installed must meet the requirements shown in drawings approved under theMaster Drawing List No. DL-206/407MM-110 revision N/C dated 27 July 1999 or later approved revision.On Bell 206B model (...) | 21.08.2007 |
| 10016956 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01352 | | FAA STC SR00486DE | BELL 407 | Installation of mounting hardware and approval for the aircraft to be operated by a single pilot from the left crew seat. | | DL-407LS-100 revision N/C dated 1 April 2002 or later approved Revision # Master Drawing List.PR-407LS-100M revision 1 dated 3 March 2003 or later approved revision # Rotorcraft Flight Manual Supplement.PR-407LS-120M revision 0 dated 5 December 2002 (...) | For left seat operation of equipment originally intended tobe operated by the pilot from the right crew seat, provisions must be made to ensure that such equipment can be equallyoperable from the left crew seat with similar controls. | 21.08.2007 |
| 10016957 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01353 | | FAA STC SR00411DE | BELL 407 | Installation of Mounting Hardware to Support Blade Folding Kit. | | DL-407BF-100 revision N/C dated 30 November 1999 or later approved Revision - Master Drawing List.PR-407BF-120 revision N/C dated 2 February 2000 or later approved Revision - Instructions for Continued Airworthiness. | | 10.09.2007 |
| 10016964 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01365 | | FAA STC SR00574DE | BELL 407 | Installation of 407FL-100 FADEC/Check instrument auxiliary annunciation system. (This STC has been classified as #BASIC# by the FAAin accordance with FAA Order 8110.52 and has been approved using the PTVP Procedures.) | | DL-407FL-100 revision A dated 7 September 2004 or later approved Revision # Master Drawing List.PR-407FL-100M revision Original dated 26 January 2005 or later approved revision # Rotorcraft Flight Manual Supplement.PR-407FL-120M revision 1dated 7 Se (...) | | 06.09.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------|-------------------|----------------|----------------------|---|---|--|--|--|---------------|
| 10016965 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01366 | REV. 1 | FAA STC SR00349DE | BELL 407 | Installation of a Searchlight. This STC has been classified as #BASIC# by the FAA and has been approved under the PTVF. | | DL-407SX-100 revision A dated 27 April 1998 or later approved revision # Master Document ListPR-407SX-100M revision 0 dated 26 June 1998 or later approved revision # Flight manual Supplement. | | 19.03.2008 |
| 10016966 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01367 | | FAA STC SR00348DE | BELL 407 | Infrared Imaging System Installation. This STC has been classified as #BASIC# by the FAA and has been approved under the PTVF. | | DL-407IR-100 revision A dated 21 May 1998 or later approved revision # Master Document ListPR-407IR-100M revision 4 dated 8 July 2005 or later approved revision # Rotorcraft Flight Manual Supplement. | None. | 22.05.2008 |
| 10016998 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01408 | | FAA STC SH2726NM | 269A, 269B, 269C | Cabin Heater Installation. | | Drawing list DL-269H, rev. D, or later approved revision Flight Manual Supplement dated November 22, 1985, or later approved revision | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 24.09.2008 |
| 10017038 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01462 | | FAA STC SR00401DE | BELL 206L/L1/L3/L4 AND 407 | Snow deflector Installation. | | DL-206/407SD-200 revision C Dated 8 January 2007 or later approved revision # Drawing List;PR-206/407SD-120M revision 3 dated 12 December 2005 or later approved revision # Instructions for Continued Airworthiness | | 18.07.2008 |
| 10017041 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01465 | | FAA STC SR00334DE | BELL 407 | Pneumatic Door Opener (PDO). This STC has been classified as #BASIC# by the FAA and has been approved under the PTVF. | | DL-407PDO-100 revision (-) dated 22 October 1997 or later approved revision # Master Drawing List;PR-407PDO-120M revision 1 dated 12 December 2005 or later approved revision # Instructions for Continued Airworthiness | | 09.06.2008 |
| 10017056 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01487 | | | BELL 206, BELL 407 | Vent Control Certification. | | 206VC-100 revision NC dated 22 November 2006 or later approved revision # Master Drawing List;PR-206VC-120M revision 0 dated 22 September 2006 or later approved revision # Instructions for Continued Airworthiness. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 11.08.2008 |
| 10017061 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01496 | | FAA STC SR00654DE | 206L-3, 206L-4, BELL 407 BELL 206A, 206B, 206L, 206L-1, | Rotary Latch Belts Installation. | | DL-407SB-100 revision A dated 1 November 2007 or later approved revision # Master Drawing List;PR-407SB-120M revision 1 dated 2 November 2007 or later approved revision # Instructions for Continued Airworthiness | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 03.09.2008 |
| 10017064 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01500 | | FAA STC SR00532DE | BELL 212, ,412, 412EP | ENHANCED VISION INSTALLATION (This STC has been classified as #BASIC# by the FAA in accordance with FAA Order 8110.52 and has been approved using the PTVF Procedures.) | | DL-212/412EV-100 revision G dated 11 July 2008 or later EASA approved revision # Master Data List;PR-212/412EV-120M revision 3 dated 7 May 2007 or later EASA approved revision # Instructions for Continued Airworthiness;PR-212/412EV-100M revision 1 d (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 23.09.2008 |
| 10017065 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01501 | | FAA STC SR00603DE | BELL 206, BELL 407 | Enhanced Vision Installation. | | DL-206/407EV-100 revision N/C dated 10 April 2006 or later EASA approved revision # Master Drawing List;PR-206/407EV-100M revision N/C dated 28 July 2006 or later EASA approved revision # Flight Manual Supplement;PR-206/407EV-120M revision 0 dated 10 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 23.09.2008 |
| 10039011 | PARAVION TECHNOLOGY INC. | | | FAA STC SR00743DE | BELL 429 | Installation of structural provisions for a blade fold kit in accordance with Paravion Technology Master Drawing List, Report No DL-429BF-100, Revision A, dated December 30, 2011. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Paravion Technology Master Drawing List, Report No DL-429BF-100, Rev.A, dated December 30, 2011;- ICA Paravion Technology Document No PR-429BF-120M, Rev 0, dated September 06, 2011; or later revisions of the above listed documents approved by EASA in (...) | None. | 29.03.2012 |
| 10043067 | PARAVION TECHNOLOGY INC. | EASA.IM.R.S.01533 | | | A109 A/-C/-E/-K2/S, A119 AW119MKII | Installation for fixed receivers for ground handling blade folding kit | | - PR-A109BF1, Rev. N/C dated 18 June 2008- PR-A109BF-6, Rev. N/C dated 16 July 2008- PR-A109BF-2, Rev. N/C dated 16 July 2008- PR-A109BF-5, Rev. N/C dated 21 July 2008- DL-A209BF-100, Rev. A dated 17 November 2008- PR-A109BF-4, Rev. N/C dated 21 Jul (...) | none | 11.05.2009 |
| 10041121 | PARKER HANNIFIN | | | FAA STC SH363NE | AS 365N2, AS 365N3 SA 365N, SA 365N1 | Installation of Parker 1C44-2 fuel pump in accordance with installation Manual SP1C44-2. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Instructions Airborne SP1C44-2 Fuel Boost Pump, revision B, dated August 17, 2000. | None. | 22.08.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|---------------------|----------------|-------------------------|---|--|---|---|--|---------------|
| 10041145 | PARKER HANNIFIN | | | FAA STC SH362NE | AS355E, AS355F, AS355F1, AS355F2 | Installation of Parker 1C44-2 Fuel Pump | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Instructions for 1C44-2 Centrifugal Fuel Boost Pump -revision B dated 17 August 2000. | None | 23.08.2012 |
| 10041148 | PARKER HANNIFIN | | | FAA STC SR153CH | AS 332 L, AS 332 L1 | Installation of Parker 1C64-15 Fuel Pump | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Instructions for 1C64-15 Centrifugal Fuel Boost Pump -revision - dated 25 March 2002. | None | 24.08.2012 |
| 10027890 | PARKER HANNIFIN CORPORATION | P-EASA.IM.A.S.03272 | REV. 1 | FAA STC SA01376CH | PC-12, PC-12/45, PC-12/47, PC-12/47E | EASA Validation of STC SA01376CH | CS-23.B NSC | Airplane Flight Manual Supplement AFMS199-241 Airplane Maintenance Manual Supplement AMMS199-241 Installation Manual IM 199-241 Component Maintenance Manuals: CM 30 244 / CM 40-424 | None | 04.12.2009 |
| 10039114 | PARKER HANNIFIN CORPORATION | | | FAA STC SA619GL | 1. C90A, C90GT, C90GT1 2. 65-90, 65-A90 2. 65-A90-1 (JU-21A) 2. 65-A90-4 (RU-21E, RU-21H) 2. B90, C90, E90, H90 (T-44A) | EASA Validation of FAA STC SA619GL Installation of Parker Hannifin Main Wheel and Brake Conversion Kit Part Numbers 199-90 in accordance with Parker Hannifin Conversion Kit Installation Manual IM 199-90, Revision J, dated January 7, 2009, or later FAA approved revisions. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Parker Hannifin Conversion Kit Installation Manual IM 199-90, Rev. J, dated January 7, 2009. Instructions for Continued Airworthiness of the Main Wheel and Main Brake in accordance with the Parker Hannifin Conversion Kit 199-90 or later revisions of th (...) | None | 10.04.2012 |
| 10043296 | PARKER HANNIFIN CORPORATION | | | FAA STC SA890GL | 1. B200, B200C, B200GT, B200CCT 2. 200, 200C, 200CT, 200T, A200 2. A200C, A200CT, B200CT, B200T | Installation of Parker Hannifin Corporation, Aircraft Wheel & Brake Division Conversion Kit 199-110, revision AB, dated June 18, 2010, and if applicable, installation of Parker Hannifin Corporation, Aircraft Wheel & Brake Division Hot Air Manifold Kit 199-168, revision F, dated June 18, 2010. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Installation Drawing 50-79, revision N dated June 18, 2010; Continued Airworthiness in accordance with Parker Hannifin Corporation, Component Maintenance Manual CM30-146, revision D, dated January 19, 2011 and Component M (...) | This installation is restricted to those aircraft with standard landing gear (1&5.5 Main Wheel & Brakes). | 18.01.2013 |
| 10033420 | PARMA - TECHNIK, S.R.O. | | | | 1. M 137A, M 137AZ 2. M 332, M 332A, M 332AK 3. M 337, M 337A, M 337AK | Prolongation of TBO for the M137, M332 and M337 engines in service according to the procedure No. WMS - OS/09/04 (Definition document) under the AP DOA Parma Technik organization No. M332, 137, 337/001 | The Certification Basis for the original product remains applicable to this certificate/ approval. | Walter-Motor-Service procedure No.WMS-OS-09/04, Rev.2, 27.10.2010 | none | 19.01.2011 |
| 10027222 | PATS AIRCRAFT L.L.C. | P-EASA.IM.A.S.03142 | REV. 1 | FAA STC ST00365NY-D | CL-600-2B19 | Revisions to Auxiliary Fuel System Structural Provisions | IM.A.023 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CRI H-01 The requirements for environmental protection and the associated certificated (...) | EASA STC No. 10027222 dated 15 September 2009. FAA STC No. ST00365NY-D dated 14 January 2011. Configuration 6. Installation of a two tank Auxiliary Fuel System within the fuselage pressure vessel forward of the baggage compartment in accordance with PAT (...) | PATS Aircraft, LLC. Airplane Flight Manual Supplement No. 200, Revision B, FAA approved 11 June 2009, is required with configurations 5 and 6. | 12.07.2012 |
| 10027223 | PATS AIRCRAFT L.L.C. | P-EASA.IM.A.S.03235 | REV. 2 | FAA STC ST01730NY-D | CL-600-2B19 | Installation of an Executive Interior | EASA IM.A.023 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CRI H-01 The requirements for environmental protection and the associated certificated (...) | Configuration 1: EASA STC No. 10027223 Revision 1 dated 08 February 2010. FAA STC No. ST01730NY-D dated 14 January 2011. PATS Aircraft, LLC Master Drawing List 6049-001-MDL, revision F dated 01 July 2009. PATS Aircraft, LLC. Airplane Flight Manual Suppl (...) | In this installation the maximum passenger seating capacity is limited to 15. | 12.07.2012 |
| 10037491 | PATS AIRCRAFT L.L.C. | | | FAA STC ST01725NY-D | B737-900ER | Installation of ten auxiliary fuel tank system configurations/Validation of STC ST01725NY-D Approved configurations are the following: Configuration 1: „Installation of a Five-Cell Auxiliary Fuel System in the forward and aft cargo compartments in accordance (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this approval: CRI H-1 issue 3 dated 30/11/2011 CS 25.981 issue 6 The certificated noise and/or emissions levels o (...) | Document TO-3868 revision IR dated 29 November 2011 #Boeing 737-900ER (BBJ3) index of applicable FARs and JARs (STC ST01725NY-D Validation)- Certification plan and System Description TO-4686 revision A dated November 17, 2011- Certification plan and (...) | Configurations 1 & 5; PATS Aircraft, LLC. Airplane Flight Manual Supplement No. 221 for a Five-Cell (1 Fwd/4 Aft) Auxiliary Fuel System, revision IR dated January 30, 2009 is required. Approved for aircraft with NGS. Configurations 2 & 6; PATS Aircraft (...) | 12.12.2011 |
| 10042981 | PATS AIRCRAFT L.L.C. | EASA.IM.A.S.01751 | | FAA STC ST00936NY-D | B737-700IGW | Installation of Auxiliary Fuel System Configurations (Configurations 1-4 through 4-K in accordance to STC ST00936NY-D inclusive amendment dated June 6, 2006 EASA Certification Basis: specified in TCDS JAA25-97-018 plus INT/POL/25/12 Fuel Tank Safety, see also LBA EMZ TA0707 and CAA UKB737-800 CRI E-01 | | PATS Aircraft, LLC Airplane Flight Manual Supplement in accordance to FAA STC ST00936NY-D amended January 3, 2006; PATS Aircraft, LLC Aircraft Maintenance Manual Supplement No. TO-667. (Supplemental structural inspection requirements specified in Sect (...) | 1. The installation of the auxiliary fuel tank system, installed by this supplemental type certificate, has been evaluated in accordance with AC120-42A, Para (8). This installation has no adverse effect on previously approved ETOPS capability of the aircraft (...) | 07.06.2006 |
| 10042982 | PATS AIRCRAFT L.L.C. | EASA.IM.A.S.01752 | | FAA STC NO. ST00365NY-D | CL-600-2B19 | Installation of a two tank Auxiliary Fuel System. | | -FAA STC No. ST00365NY-D dated 29 January 2007 - Configuration 3. Installation of structural provisions only, for a two(2) Cell Auxiliary Fuel System within the fuselage pressure vessel forward of the baggage compartment in accordance with PATS Aircraft (...) | 1. PATS Aircraft, LLC Airplane Flight Manual Supplement No.200, Rev. IR FAA approved January 5, 2007 is required with configuration 5.2. PATS, Inc. Aircraft Maintenance Manual Supplement No. 1, Report No. TO-417, Revision 3 dated May 23, 1995 or later (...) | 12.06.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------|--------------------|----------------|----------------------|--|--|---|--|---|---------------|
| 10016198 | PEMCO WORLD AIR SERVICES | EASA.IM.A.S.02438 | REV. 2 | FAA STC SA2968SO | BOEING 737-400 | Modification of floor in accordance with Pemco World Air Services, Inc. Master Data list, Report No. 2316, revision #K#, dated June 15, 2006, or later EASA approved revision. | | - Master Data list, Report No. 2316, revision #K#, dated June 15, 2006, or later EASA approved revision. - Instructions for Continued Airworthiness (ICA), Pemco Report No 2917, Revision C, dated January 18, 2006, or later EASA approved revision, (...) | Instructions for Continued Airworthiness (ICA), Pemco Report No 2917, Revision C, dated January 18, 2006, or later EASA approved revision, is a required part of this STC, and must be incorporated into the operator's maintenance program. | 24.06.2008 |
| 10016199 | PEMCO WORLD AIR SERVICES | EASA.IM.A.S.02439 | REV. 2 | FAA STC SA2969SO | BOEING 737-400 | Installation of a cargo door in accordance with Pemco World Air Services, Inc. Master Data list, Report No. 2373, revision #T#, dated June 15, 2006, or later EASA approved revision. | | - Master Data list, Report No. 2373, revision #T#, dated June 15, 2006, or later EASA approved revision. - Airplane Flight manual Supplement (AFMS), Report P03-003FM Rev 7, or later EASA approved revision. - Instructions for Continued Airworthiness (ICA), Pemco Report No 2917, Revision C, dated January 18, 2006, or later EASA approved revision, (...) | - Airplane Flight manual Supplement (AFMS), Report P03-003FM Rev 7, or later EASA approved revision. - Instructions for Continued Airworthiness (ICA), Pemco Report No 2917, Revision C, dated January 18, 2006, or later EASA approved revision, is (...) | 24.06.2008 |
| 10016200 | PEMCO WORLD AIR SERVICES | EASA.IM.A.S.02440 | | FAA STC SA2970SO | BOEING 737-400 | Installation of a Class #E# Cargo Compartment and passenger interior configuration in accordance with Pemco World Air Services, Inc. Master Data list, Report No. 2374, revision #S#, dated June 15, 2006, or later EASA approved revision. | | - Master Data list, Report No. 2374, revision #S#, dated June 15, 2006, or later EASA approved revision. - Airplane Flight manual Supplement (AFMS), Report P03-003FM Rev 7, dated April 27th 2007 or later EASA approved revision. - Instructions for (...) | - Airplane Flight manual Supplement (AFMS), Report P03-003FM Rev 7, dated April 27th 2007 or later EASA approved revision. - Instructions for Continued Airworthiness (ICA), Pemco Report No 2917, Revision C, dated January 18, 2006, or later EASA (...) | 07.05.2007 |
| 10035332 | PEMCO WORLD AIR SERVICES | EASA STC 2004-3750 | REV. 1 | FAA STC SA2971SO | B737-300 | Installation of a Cargo Loading System in acc with MDL 2337rev X dated 12/07/2010. Modification of the STC FAA SA2971SO to increase from 121 pax to 148 pax according to drawing 90945-507/607 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Refer to Airworthiness Approval Note 24984 Addendum 1 (enclosed) The certificated noise (...) | - Project Specific Certification Plan for up-date of STC SA2971SO to include 148 passenger configuration of the Quick Change Handling/loadingsystem for B737-300 S/N 23522 and 23523 - Report No. ER-5085 rev 3 dated 06/09/2011 - Drawing 90945-507/607 as de (...) | None | 16.06.2011 |
| 10016480 | PENTASTAR AVIATION, L.L.C. | EASA.IM.A.S.02832 | | FAA STC ST01759CH | FALCON 50 FALCON 900 FALCON 900EX | EASA Validation of FAA approved STC ST01759CH Upgrade of a Rockwell Collins SAT-906 SATCOM system by modification of Rockwell Collins SDU-906 Satellite Data Unit and HPA-901A High Power Amplifier and the installation of a Rockwell Collins HST-900 Hg (...) | | Pentastar Aviation Engineering Report No. 02ER-004, Revision IR, dated 17th December 2002. Instructions for Continued Airworthiness, Pentastar Aviation Maintenance Supplement No. 02CA-004 Original Issue dated 28th October 2002 or later revisions of this (...) | 1. See those of original FAA STC ST01759CH.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse (...) | 01.04.2009 |
| 10044906 | PENTASTAR AVIATION, L.L.C. | | | FAA ST01966CH | G-1159, G-1159A, G-1159B GIV, GV, GV-SP | Installation of provisions for dual CMC Electronics - CMA-1100 Pilot/View/TM Electronic Flight Bag (EFB) system | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - FAA STC FAA ST01966CH, reissued 14th January 2008. - Pentastar Aviation Top Level Report No. 03TL-004, Revision J, dated 10th January 2008. - FAA Approved Airplane Flight Manual Supplement No. 05AFM-017 Revision B, dated 7th October 2009. - Instructio (...) | Compatibility of this design change with previously approved modifications must be determined by the installer. The equipment for which these provisions are intended has not been certified. Additional EASA approval is required for the installation of this (...) | 16.05.2013 |
| 10013930 | PERRENOUD, DIDIER | EASA.A.S.00277 | REV. 2 | | B300 | Installation of Aerodata Automatic Flight Inspection System AD-AFIS-112. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-23, Amendment 2 (elected to comply by the applicant, not required by 21A.101) The requ (...) | Installation Manual ,,,,, HB-A/C-1087, Rev. 0. Master Document List ,,,,, MDL-A/C-067, Rev. 8. Maintenance Manual Supplement, , AMM-1088, Rev. 0. Flight Manual Supplement, , AFM-124, Rev. 1, or later revisions of the above listed documents approved by EASA. | None | 06.05.2011 |
| 10013931 | PERRENOUD, DIDIER | EASA.A.S.00278 | REV. 2 | | BEECH 300 | Flight Data Recorder (FDR) integration into Hawker Beechcraft King Air B300 | | Existing STC ref EASA.A.S.00278 covers FDR and CVR (Cockpit Voice Recorder) , this change to the STC covers integration of the FDR only, (partial installation of STC EASA.A.S.00278) Installation, Operation and Continued airworthiness in accordance with (...) | | 19.10.2007 |
| 10013988 | PERRENOUD, DIDIER | EASA.A.S.01020 | | | BEECH B200 BEECH B200T | Installation of TCAS 2000 with RC2 852 Mode S Transponders | | - AFM 075 Rev 0 or later EASA approved revision- AMM-772 Rev 0 | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.02.2005 |
| 10013989 | PERRENOUD, DIDIER | EASA.A.S.01021 | | | BEECH B200 BEECH B200T BEECH B300 | Installation of UNS 1-F Flight Management System | | - AFM 076 Rev 1 or later EASA approved revision- AMM-774 Rev 0 | | 11.02.2005 |
| 10013991 | PERRENOUD, DIDIER | EASA.A.S.01023 | | | DORNIER D328-100 DORNIER D328-300 | AeroPhone Iridium SATCOM Installation | | Master Document List, Ref MDL-A/C-062, Revision 3 dated 8 September 2005. Airplane Flight Manual Supplement, Ref AFM-058, Revision 1, dated 13 August 2005. Airplane Maintenance Manual Supplement, Ref HB-A/C-708, Revision 3 | | 14.09.2005 |
| 10013993 | PERRENOUD, DIDIER | EASA.A.S.01024 | REV.11 | | DORNIER 328-100 | Installation of Maritime Surveillance System on Dornier 328-100 including SAR Radar, FLIR, and optional IR/UV Scanner. Revision 4: Functional/optional upgrade to the radar system and mission software to optimize search and rescue operation. Hardware upgr (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Master Document List ref MDL-A/C-089 Rev. 13. - Flight Manual Supplement ref AFM-077 Rev. 10. - Aerodata Maintenance Manual Supplement AMM-782 Rev. 9 dated September 2012 or later revisions of the above listed documents approved by EASA | Applicable to Aircraft Serials 3051, 3052, 3053, 3057 and 3059 as modified by Aerodata Modification D328-2.xx for AMSA Mission Purpose. Prior to installation of this design change it must be determined that the interrelationship between this design cha (...) | 29.01.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------|----------------|----------------|----------------------|--|---|--|---|--|---------------|
| 10014244 | PERRENOUD, DIDIER | EASA.A.S.01454 | | | BEECH B200 SUPER KING AIR | Installation of Aerodata Flight Inspection System AD-FIS-9 | | - Master Document List MDL-A/C-100 Rev. 1 EASA approved Airplane Flight Manual Supplement for Flight Inspection System, Model AD-FIS-9, in Beech King Air B200, Doc. No. AFM-081 Rev. 0; Airplane Maintenance Manual Supplement AMM-799 | | 25.01.2006 |
| 10014515 | PERRENOUD, DIDIER | EASA.A.S.02591 | | | | Installation of Aerodata Flight Inspection System AD-AFIS-0220. | | Installation in accordance with Master Document List MDL-A/C-110 Rev 1 or later revisions. Operation in accordance with Flight Manual Supplement AFM-082 Rev 0, or later approved revisions. Instructions for Continued Airworthiness AMM-516, Rev 0, or later approved revisions. | | 31.10.2006 |
| 10014516 | PERRENOUD, DIDIER | EASA.A.S.02592 | | LBA STC SA 1320 | BEECH B200 | Installation of Normarc Flight Inspection System NM36258. NOTE: this STC supersedes existing LBA STC SA 1320 | | Installation in accordance with Master Document Listref MDL-A/C-059 Rev 4 or later approved revisions. Operation in accordance with Flight Manual Supplementref FMS-056 Rev 1, or later approved revisions. Instructions for Continued Airworthinessref HB-A/C-676, Rev 3, or later approved revisions. | None | 07.09.2006 |
| 10014785 | PERRENOUD, DIDIER | EASA.A.S.02985 | | | BEECH B300 | Installation of EGWS Mark VI Upgrade in B300. | | Installation in accordance with Master Document Listref MDL-A/C-117 Rev 3 or later revisions. Operation in accordance with Flight Manual Supplementref AFM-084 Rev 0, or later revisions. Instructions for Continued Airworthinessref AMM-858, Rev 0, or later revisions. | Compatibility of this modification with other previously approved modifications must be determined by the installer. | 16.04.2007 |
| 10014930 | PERRENOUD, DIDIER | EASA.A.S.03184 | REV. 1 | | KING AIR 300 | Installation of an Aerodata Flight Inspection System AD-AFIS-230 | | EASA approved Master Document List, Doc. No. MDL-A/C-122, Revision 1, dated 15 October 2007, or later approved revision. EASA approved MOD-Order List, Doc. No. MOL-B300-11.01, Revision 0 dated 20 July 2007, or later approved revision. EASA approved Ai (...) | Airplane must be equipped with Collins ProLine II to be eligible for this STC. | 26.10.2007 |
| 10015136 | PERRENOUD, DIDIER | EASA.A.S.03489 | REV. 3 | | B200, B200C, B200GT | Installation of Aerodata Flight Inspection System- AD-AFIS-120, Mod. No. B200-11.01, or- AD-AFIS-250, Mod. No. B200-12.01, or- AD-AFIS-250, Mod. No. B200-12.02, or- AD-AFIS-250, Mod. No. B200-14.01 | Design: Non-significant change The Certification Basis of the original product and the following additional / alternative airworthiness requirements are applicable to this change: CS-23, Amdt. 1, except 23.561 (kept FAR 23.561 at Amdt. 7) is elected tocom (...) | For AD-AFIS-120, Mod. No. B200-11.01: EASA approved Master Document List, Installation of Aerodata FIS, Doc. No. MDL-A/C-129, Rev. 3, dated 27-Nov-2009, or later approved revision. EASA approved Airplane Flight Manual Supplement, Doc. No. AFM-094, Rev.4 (...) | For AD-AFIS-120, Mod. No. B200-11.01: Only applicable for aircraft model B200 and B200GT, and when equipped with Collins Proline 21 <(->-) and preconditions as defined in Design Description, Installation of Aerodata FIS, Doc. No. DES-A/C-836, Rev. 2, dat (...) | 08.06.2011 |
| 10015166 | PERRENOUD, DIDIER | EASA.A.S.03536 | REV. 1 | | CESSNA 550 CITATION II | Additional Cockpit Information Display The current STC EASA.A.S.03536 covers the installation of an Aerodata Flight Inspection System (FIS). To further reduce the required verbal communication and therefore the workload of the cockpit crew and the FI (...) | | Master Document List MDL-A/C-137 Revision 2 Airplane Flight Manual Supplement AFM-095 Revision 1 Airplane Maintenance Manual Supplement AMM-952 Revision 1 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 26.01.2009 |
| 10015182 | PERRENOUD, DIDIER | EASA.A.S.03561 | | | PILATUS PC-12 | Modification as surveillance aircraft | | 1. According to Aerodata Master Document List Doc. No. MDL-A/C-138 Rev. 1, or later Revisions. 2. To be operated in accordance with Aerodata AFM-S Document No. AFM-100, Rev. 0, or later approved Revisions. 3. To be maintained i.a.w. Aerodata A (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 11.11.2008 |
| 10015235 | PERRENOUD, DIDIER | EASA.A.S.03660 | | | LEARJET 31A | Installation of Aerodata Flight Inspection System (FIS) | | As detailed in Master Document List MDL-A/C-139, rev. 1 (or later approved revisions) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 16.10.2008 |
| 10015270 | PERRENOUD, DIDIER | EASA.A.S.03745 | | | BEECH 300 | Installation of an Aerodata Flight Inspection System AD-AFIS-111 | | 2 EASA approved Master Document List, Doc. No. MDL-A/C-144, Revision 1, dated 10-Mar-2009, or later approved revision; EASA approved Airplane Flight Manual Supplement, Doc. No. AFM 104, Revision 0, dated 20-Nov-2008, or later approved revision (...) | 1. Airplane must be equipped as stated in Chapter 1.3.1 of EASA approved Aerodata Certification Program, Doc. No. CP-B300-13.01, Revision 0, dated 12-Jan-2009, or later approved revision. 2. Prior to installation of this modification the installer (...) | 13.03.2009 |
| 10017098 | PERRENOUD, DIDIER | EASA.R.S.01022 | REV. 1 | | EUROCOPTER DEUTSCHLAND BK117 B EUROCOPTER DEUTSCHLAND BK117 C | - Cockpit Arrangement for ADAC BK 117 Fleet for IFR operation- Re-arrangement of Cockpit Instruments for a harmonized Cockpit Layout of several BK 117 Helicopters. Furthermore integration of a dual Garmin GNS430 equipment and GTX330 Mode S Transponder | | - Master Document List MDL-A/C-092 Rev. 6 or later- BK117 Flight Manual Supplement #Cockpit Modification BK117 Modification Number BK117-2.054, AFM-078 Rev. 1- Instructions for continued airworthiness #Avi (...) | 1 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 10.02.2009 |
| 10017120 | PERRENOUD, DIDIER | EASA.R.S.01205 | | | EUROCOPTER DEUTSCHLAND MBB-BK | Installation of AeroPhone IRIDIUM SATCOM Installation of satellite telephone system AeroPhone on helicopter | | Master Document List MDL-A/C-093 Rev. 1 or later Installation and Testing Instructions REP-A/C-1258 Rev. 0 or later Flight Manual Supplement AFM-079 Rev. 0 or later Continued Airworthiness Instructions AMM-791 Rev. 0 or later | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer | 31.08.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------|------------------|----------------|----------------------|--|--|--|---|--|---------------|
| 10017257 | PERRENOUD, DIDIER | EASA.R.S.01464 | | | EUROCOPTER DEUTSCHLAND EC-135- | Airborne observation platform (AOP) Integration in EC135. Installation of a Star Safire HD FLIR on the right skid and an operator station in the passenger compartment based on Aerodata change no. EC135-3.02 iaw MDL/A/C-132 Rev. 1 or later approved revisions. | | Flight Manual Supplement AFM-096, Rev. 1. Maintenance Manual Supplement AMM-956, Rev. 0. Installation Instruction HB-A/C-957, Rev. 1. | None. | 29.08.2008 |
| 10017258 | PERRENOUD, DIDIER | EASA.R.S.01465 | | | EUROCOPTER FRANCE EC 155 B1 EUROCOPTER FRANCE EC 155B | Airborne observation platform (AOP) Integration in EC155. Optional Installation of a operator station and an external FLIR camera system as replacement for an existing equipment. The installation is in accordance with Aerodata Major change EC155-3.01, (...) | | 2 Rotorcraft Flight Manual Supplement #AOP Integration in EC155# AFM-107 Rev. 1 or later approved revision. Instructions for continued airworthiness: #Maintenance Manual Supplement Installation of Aerodata AD-MS-0100 in EC155 helicopter#, AMM- (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 18.03.2009 |
| 10017277 | PERRENOUD, DIDIER | EASA.R.S.01523 | | | EUROCOPTER DEUTSCHLAND EC-135- | 1. Title: Installation of AeroPhone IRIDIUM SATCOM in EC135.2. Description: Installation of an AeroPhone IRIDIUM SATCOM system in EC135 based on AERODATA change no. AD-IRI-3.02 i.a.w. the MDL MDL/A/C-093 Rev.3 or later approved revision. | | 1. Flight Manual Supplement: # Flight Manual Supplement AFM-079 Rev. 12. Installation Instruction, Instructions for Continued Airworthiness: # Maintenance Manual Supplement AMM-791 Rev. 0# Installation and Testing Instructions REP-A/C-1258 Rev. 2 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 13.11.2008 |
| 10027409 | PERRENOUD, DIDIER | P-EASA.A.S.03795 | | | BEECH B300 | Installation of Aerodata Flight Inspection System AD-AFIS-150 | The following new Certification Basis is applicable to this change ("ICS-23, Amdt. 1, is elected to comply with for the items affected by the change. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | EASA approved Master Document List, Doc. No. MDL/A/C-145, Rev. 1, dated 25-Aug-2009, or later approved revision. EASA approved Airplane Flight Manual Supplement, Doc. No. AFM-106, Rev. 1, dated 25-Aug-2009, or later approved revision. Approved Maintena (...) | Only applicable for aircraft / preconditions as defined in Design Description, Doc. No. DES-A/C-884, Rev. 1, dated 03-Aug-2009, or later approved revision | 29.09.2009 |
| 10027987 | PERRENOUD, DIDIER | P-EASA.A.S.03960 | | | B300 | Installation of Aerodata Flight Inspection System AD-AFIS-260 | The following new Certification Basis is applicable to this change ("CS-23, Amdt. 1, except 23.561 and 23.785 (kept FAR 23.561 and FAR 23.785) is elected to comply with for the items affected by the change. The certified noise and/or emissions levels of (...) | EASA approved Master Document List, Integration of AD-AFIS-260, Doc. No. MDL/A/C-149, Rev. 1, dated 13-Nov-2009, or later approved revision. EASA approved Airplane Flight Manual Supplement, Doc. No. AFM-108, Rev. 0, dated 06-Oct-2009, or later approved r (...) | Only applicable for aircraft / preconditions as defined in Design Description, Integration of AD-AFIS-260, Doc. No. DES-A/C-887, Rev. 2, dated 19-Oct-2009, or later approved revision | 23.11.2009 |
| 10029093 | PERRENOUD, DIDIER | P-EASA.A.S.03846 | | | 1329-25 (JETSTAR II) | Avionics Upgrade for RVSM Compatibility. Design definition: Aerodata's documents reference MDL-A/C-075 Rev. 3, 23/02/2010, Master Document List | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: As defined in "Lockheed 1329-25 Jetstar II Avionics Upgrade for RVSM Compatibility (P-EA (...) | Aerodata's documents reference: -AFM-069 Rev. 2, 11/02/2010, EASA approved supplement to Flight Manual and -AMM-748 Rev. 2, 22/02/2010, supplement to Maintenance Manual, including EASA approved Chapter 5, "Time Limits/Maintenance Checks" or later revisions of the above listed documents approved by EASA | Limited to: S/N 5226 | 03.03.2010 |
| 10030547 | PERRENOUD, DIDIER | P-EASA.A.S.03903 | | | L-410UVP-E15 | Installation of Aerodata Flight Inspection System AD-AFIS-130 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: EASA CS-23 and FAR Part 23 Arndt. 34, §23.561 & 23.785 (chosen by applicant). The require (...) | Airplane Flight Manual Supplement (AFM-110 Rev 0) Airplane Maintenance Manual (AMM-1010 Rev 0) or later revisions of the above listed documents approved by EASA | Limited to serial no. 892318 and 892329 | 23.06.2010 |
| 10033638 | PERRENOUD, DIDIER | | REV. 1 | | BAE.125 SERIES 800A HAWKER 800 (U125) | Installation of Aerodata FIS (Flight Inspection System). The FIS installation consists of non-essential sensor systems in the cabin and a camera system in the tail cone to enable calibration missions for navigation equipment. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Aerodata Master document list, MDL/A/C-155, Rev. 2. Operation in accordance with Aerodata Flight Manual Supplement, AFM-113, Rev. 1. Continued Airworthiness in accordance with Aerodata Aircraft Maintenance Manual Supplement, (...) | none | 25.02.2011 |
| 10034248 | PERRENOUD, DIDIER | | REV. 1 | | B200 | Modification for Maritime Patrol Mission. The modification consist of the installation of a belly mounted radome with integrated camera elevator, a drop hatch and bubble windows designed by Hawker Beechcraft and the integration of non-essential systems a (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certified noise and/or emissions levels. | Installation in accordance with Aerodata Master Document List MDL/A/C-157, Rev 3, dated 4. March 2011. - Operation in accordance with Aerodata Supplemental Airplane Flight Manual, AFM-116, Rev. 1 dated 4. March, 2011. - Continued Airworthiness in accordance (...) | None | 18.09.2012 |
| 10036103 | PERRENOUD, DIDIER | | REV. 1 | | B300 | Installation of Aerodata Automatic Flight Inspection System AD-AFIS-112. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23, Amendment 2. The requirements for environmental protection and the associated (...) | Flight Manual Supplement: AFM-126, Rev. 1- Installation Manual HB-A/C-1101, Rev. 0- Master Document List MDL/A/C-0161, Rev. 2- Maintenance Manual Supplement AMM-1099, Rev. 1 or later revisions of the above listed documents approved by EASA. | None. | 16.08.2011 |
| 10036478 | PERRENOUD, DIDIER | | | LBA EMZ NR. TA0607 | FALCON 10 | Change to LBA STC approval "EMZ Nr. TA0607 and AFM approval DTM-721A" dated 26 June 2002. Change of procedures "Instructions for Continued Airworthiness" in relation with RVSM qualification and update of the aircraft configuration. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: JAR 25 Change 15 and JAA TGL 6. The requirements for environmental protection and the ass (...) | Master Data List doc. N° MDL/A/C-047, Rev. 5 dated September 08 2011. Instructions for Initial and Continued Airworthiness doc. N° HB-A/C-650, Rev. 4 dated September 06 2011. Airplane Flight Manual Supplement doc. N° FMS-047, Rev. 2 dated August 02 2011. o (...) | Restricted to S/N 183 | 12.09.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|---------------------|----------------|----------------------|---|---|---|--|---|---------------|
| 10036479 | PERRENOUD, DIDIER | | | LBA EMZ NR. TA0600 | FALCON 10 | Change to LBA STC approval "EMZ Nr. TA0600 and AFM approval DTM-721A" dated 06 June 2002. Change of procedures "Instructions for Continued Airworthiness" in relation with RVSM qualification and update of the aircraft configuration. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:JAR 25 Change 15 and IAA TGL. The requirements for environmental protection and the ass (...) | Master Data List doc. N° MDL-A/C-044, Rev. 3 dated 06 September 2011. Instructions for Initial and Continued Airworthiness doc. N° HB-A/C-545, Rev. 2 dated 06 September 2011. Airplane Flight Manual Supplement doc. N° FMS-046, Rev. 3 dated 02 August 2011. o (...) | Restricted to S/N 216. | 12.09.2011 |
| 10039634 | PERRENOUD, DIDIER | | | | B300 | Installation of Aerodata Automatic Flight Inspection System AD-AFIS-0270 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-23, Amendment 2 (not required by 21A.101) The requirements for environmental protecti (...) | Flight Manual-Supplement: AFM-130, Rev. 1 or later revisions of the above listed documents approved by EASA. Installation Manual, HB-A/C-1118, Rev. 0 Master Document List, MDL-A/C-0177, Rev. 3 Maintenance Manual Supplement, AMM-1117, Rev. 0 | None | 15.05.2012 |
| 10040364 | PERRENOUD, DIDIER | | | | B300 | Installation of Aerodata Automatic Flight Inspection System AD-AFIS-0280 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-23, Amendment 2 The requirements for environmental protection and the associated certi (...) | Flight Manual Supplement: AFM-121, Rev. 2 or later revisions of the above listed documents approved by EASA- Installation Manual HB-A/C-1090, Rev. 0- Master Document List MDL-A/C-0170, Rev. 2- Maintenance Manual Supplement AMM-1125, Rev. 0 | none | 28.06.2012 |
| 10043030 | PERRENOUD, DIDIER | EASA.A.S.02760 | | | 650 | Installation of TCAS 2000 System including Mode S Transponders | | MDL-A/C-074 Rev.5 Master Document List AFM-068 Rev. 2 Airplane Flight Manual Supplement HB-A/C-747 Maintenance Manual Supplement | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate sha (...) | 20.07.2006 |
| 10043031 | PERRENOUD, DIDIER | EASA.A.S.00217 | | | 560 | Installation of a TCAS 2000 in accordance with Master Drawing List Doc.No. MDL-A/C-081 (->,-<)- Rev. 01 (or later EASA approved revision). | | | 1- Operation Instruction: Aero Data Flight Manual Supplement AFM-071, Rev. 02- Instruction for Continuous Airworthiness Doc. No AMM-758, Rev. 03- Installation Instruction: Modification Order Doc. No. HB-A/C-7594- This STC is approved only for the produc (...) | 06.10.2004 |
| 10043842 | PERRENOUD, DIDIER | | | | HAWKER 750 | Installation of AD-AFIS-355 consisting of operator console and equipment console for Flight Inspection System (FIS) with related antenna installations and installation of camera tube (GHT) in the tail cone. Along with associated modification to cockpit and electrical interfaces. | The Certification Basis applicable to this certificate/ approval is defined by:- CS 25 Amdt. 11 excluding Subpart H and with the following revisions:- 25.561 FAR-25 Amdt. 0 as defined on TCDS IM.A.085-25.785 FAR-25 Amdt. 20 as defined on TCDS IM.A.085-2 (...) | Installation of Aerodata AD-FIS-355 under modification H125-3.01 as defined in Aerodata MDL-A/C-185 Rev 2 dated 15 February 2013. Operation in accordance with Aerodata Flight Manual Supplement AFM-132 Rev 0 dated February 2013. Maintenance in accordance (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the produc (...) | 28.02.2013 |
| 10015853 | PETERSEN AVIATION, INC. | EASA.IM.A.S.01956 | | FAA STC SA00328W1 | REIMS AVIATION CESSNA F-172N | Operation of airplane on leaded and unleaded automotive gasoline, 91 minimum antiknock index (RON + MON) / 2 per ASTM Specification D-439. Intermixing with aviation gasoline also approved. | | Petersen Drawing No. RF-172-91 dated January 18, 1995. | Limited to those airplanes equipped with gravity fuel feed to the carburetor and STC certificated or approved engine installation. All engines must have separate approval for operation on automotive gasoline in the form of Supplemental or Type Certi (...) | 24.04.2006 |
| 10015854 | PETERSEN AVIATION, INC. | EASA.IM.A.S.01957 | | FAA STC SA2600CE | CESSNA 172D CESSNA 172E CESSNA 172F CESSNA 172G CESSNA 172H CESSNA 172I CESSNA 172K CESSNA 172L CESSNA 172M CESSNA 172N CESSNA 172P | Operation of airplane on leaded and unleaded automotive gasoline, 91 minimum antiknock index (RON + MON) / 2 per ASTM Specification D-439. Intermixing with aviation gasoline also approved. | | 1. Petersen Drawing No. 172-91 dated February 13, or later approved revision. 2. Airplane Flight Manual Supplement dated March 30, 1990, or later approved revision for 172N, Serial Number 17270050 & 17271035 thru 17274009 & 172P Models only. | Limited to those airplanes equipped with gravity fuel feed wing fuel tanks as the main fuel supply to the carburetor and certificated (original or STC) or approved engine installations. All engines must have separate approval for operation on automa (...) | 20.04.2006 |
| 10016291 | PETERSEN AVIATION, INC. | EASA.IM.A.S.02586 | | FAA STC SA2660CE | PA-28-160, PA-28-161, PA-28-180, PA-28-181 | Operation of Piper airplanes with automotive gasoline in accordance with FAA STC SA2660CE, dated April 29, 2005. | | # Airplane Flight Manual Supplement No.2, Revision B, dated July 25, 2007.# Drawing List 28-TL-2, Rev. (F), dated April 22, 1991.# Drawing List 28-TL-3, Rev. (X), dated July 25, 2007. | 1. Limitations as established in FAA STC SA2660CE, dated April 29, 2005 are applicable. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously inst (...) | 10.09.2007 |
| 10016690 | PETERSEN AVIATION, INC. | EASA.IM.E.S.01002 | | FAA STC SE00663W1 | LYCOMING ENGINES O-540-F1B5 | Add the following approved fuel: Leaded and unleaded automotive gasoline, 91 minimum antiknock index (RON+MON)/2 per ASTM Specification D-439, D-4814 or per European Norm Standard EN228. Intermixing with aviation gasoline is also approved. | | Petersen Drawing No. 540-2 dated July 10, 1990, or later approved revisions | 1. This STC is limited to engine installations on Robinson Model R44 Helicopters only. The manifold pressure and RPM limitations in the FAA Approved Robinson R44 Flight Manual must be maintained. Maximum Take off power is limited to 225 BHP. 2. If the h (...) | 23.02.2006 |
| 10030424 | PETERSEN AVIATION, INC. | P-EASA.IM.A.S.03224 | | FAA STC SA00214W1 | CESSNA F182P | Operation of airplane on unleaded automotive gasoline, 87 minimum antiknock index and leaded automotive gasoline, 88 minimum antiknock index (RON + MON) / 2 per ASTM Specification D-439 in acc. to FAA STCSA00214W1. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | EASA approved AFMS for Reims-Cessna F182 P, Doc.No. 182P-SA00214W1, Issue 1, dated June 10, 2010. Petersen Drawing No. C718984 dated July 1, 1984, revised December 15, 1992. or later revisions of the above listed documents approved by EASA | 1. Limited to those airplanes equipped with gravity fuel feed to the carburetor and certificated (original or STC) EASA Approved engine installations. All engines must have separate approval for operation on automotive gasoline in the form of Supplement (...) | 15.06.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|----------------|----------------|----------------------|-----------------------------------|--|--|---|--|---------------|
| 10033096 | PETERSEN AVIATION, INC. | | | FAA STC SA2670CE | 1. A-1, A-1A, A-1B 2. A-1C-180 | Operation of airplanes using automotive gasoline according US STCSA2670CE | The Certification Basis for the original product remains applicable to this certificate/ approval. FAR 23 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and (...) | Installed law Petersen Drawing No. A-1, dated 12 Nov 1990, FAA approved AFM-S No 1, dated 27 Nov 1990 or FAA approved AFM-S No 2, Rev. (), dated 4 Apr. 2010, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Deci (...) | -According limitation section of AFM-S- Limited to those airplanes equipped with gravity fuel feed to carburetor only- Not approved for 82UL or mixtures with 82UL- Limitation as indicated in FAA STC SA2670CE also apply | 17.12.2010 |
| 10040630 | PFW ENGINEERING DO GmbH | | REV. 1 | | 319-100 | General Fin Camera Installation on Airbus A319 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Original certification basis for A319-100 as per TCDS's EASA.A.064 issue 07 plus CS 25. (...) | GCT's documents:- Master Drawing List A319A-MDR-233000 Issue A, dated 12th July 2012 and- Master Equipment List A319A-MEQ-233000 Issue A, dated 31st March 2012, and- STC Operators Manual A319A-COM-B, Issue 30th June 2012, comprising- Instructions for (...) | This design change approval is applicable on Aircraft model Airbus A319-100, aircraft serial number 4428. | 11.10.2013 |
| 10040680 | PFW ENGINEERING DO GmbH | | REV. 2 | | 747-8 | Structural provisions for 747-8i belly camera installation between the NLG bay and the belly fairing at the lowest point of the fuselage. It contains:- Attachment to the aircraft structure through brackets- Outside fairing- Installation of awiring feed-through | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | STC Description Document DE-530-B747A Issue 01 (>-<->-) Master Document List CCN-B747A-MCDL Issue 01 (>-<->-) STC Operators Manual B747A-COM-A dated 31 Jan 2012 (>-<->-) or later revisions of the above listed documents approved by EASA. | none | 11.10.2013 |
| 10014378 | PHOENIX AEROSPACE LIMITED | EASA.A.S.01653 | | | | Installation of Dual GNS530 with Dual GI-106A & Dual KNI-582 RMI | | Phoenix Aviation Design Services Master Document List PADS/MDL/1003/001 Phoenix Aviation Design Services AFM Supplement PADS/FMS/J31-32/001 Revision 0 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 10.02.2006 |
| 10014430 | PHOENIX AEROSPACE LIMITED | EASA.A.S.02478 | | | BEECH A36 | Installation of Skywatch Traffic Avoidance | | Phoenix Aviation Design Services Modification PADS_CP_1041 Phoenix Aviation Design Services AFM Supplement PADS/FMS/436/001 Revision 0 dated March 20 2006 or later approved revision | | 07.06.2006 |
| 10014589 | PHOENIX AEROSPACE LIMITED | EASA.A.S.02696 | REV. 1 | | BN2B-26 ISLANDER | Installation of GNS530, GNS430 & GMA340. | | # Phoenix Aviation Design Services Modification PADS/CP/1057# Phoenix Aviation Design Services AFM Supplement PADS/FMS/BN2B/001 Revision 0 or later approved revision & PADS/FMS/BN2B/002 Revision 0 or later approved revision GNS530# Phoenix Aviation Design Services AFM Supplement PADS/FMS/BN | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 24.04.2007 |
| 10014732 | PHOENIX AEROSPACE LIMITED | EASA.A.S.02914 | | | CESSNA 560 | Installation of Honeywell EGPWS Cessna 560 G-PPLC ASN 560-0059 | | Phoenix Aviation Design Services Modification PADS/CP/1070. Phoenix Aviation Design Services Master Drawing List PADS/MDL/1070/001 Revision A. Phoenix Aviation Design Services AFM Supplement PADS/FMS/560/001 Revision B. | 1. Restricted to G-PPLC ASN 560-00592. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and #Associated Technical Documentation#. Compatibility with other (...) | 24.01.2007 |
| 10014979 | PHOENIX AEROSPACE LIMITED | EASA.A.S.03269 | | | | Installation of new Avionics. | | Phoenix Aviation Design Services Ltd. Change PADS/CP/1098. Flight Manual Supplements: PADS/FMS/F406/001 PADS/FMS/F406/002 PADS/FMS/F406/003 | | 30.01.2008 |
| 10015100 | PHOENIX AEROSPACE LIMITED | EASA.A.S.03443 | | | PIPER PA-31 | Installation of new Avionics | | Phoenix Aviation Design Services Limited Design Change PADS/CP/1110 Part B Flight Manual Supplements PADS/FMS/PA31/001 Revision 1 PADS/FMS/PA31/002 Revision 1 or later EASA approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 30.09.2008 |
| 10015101 | PHOENIX AEROSPACE LIMITED | EASA.A.S.03444 | | | PIPER PA-39 | Validation of Foreign Installed Avionics | | Phoenix Aviation Design Services Ltd. Major Change PADS/CP/1111 Flight Manuals Supplements PADS/FMS/PA39/001 Rev. A PADS/FMS/PA39/002 Rev. A. | | 05.05.2008 |
| 10015141 | PHOENIX AEROSPACE LIMITED | EASA.A.S.03499 | | | CESSNA 208 | Installation of Garmin GNS430 and GTX330 Mode "S" Transponder. | | Master Document List PADS/MDL/1113/001 Rev. A AFM Supplements PADS/FMS/208/001 Rev. A PADS/FMS/208/002 Rev. A. | | 19.03.2008 |
| 10015189 | PHOENIX AEROSPACE LIMITED | EASA.A.S.03573 | | | | Installation of Garmin GNS430 | | Phoenix Aviation Design Services Limited Major Change PADS/CP/1121 Flight Manual Supplement PADS/FMS/BN2/009 Rev A | | 13.06.2008 |
| 10017139 | PHOENIX AEROSPACE LIMITED | EASA.R.S.01240 | | | EUROCOPTER DEUTSCHLAND MBB-BK | Installation of Skywatch Traffic Avoidance | | Phoenix Aviation Design Services Modification PADS/CP/1040 Phoenix Aviation Design Services AFM Supplement PADS/FMS/BK117/001 Revision 0 or later approved revision | | 22.05.2006 |
| 10017180 | PHOENIX AEROSPACE LIMITED | EASA.R.S.01303 | | | EUROCOPTER FRANCE AS 365N1 | Installation for KLN90B GPS, UNS-1K Flight Management System, Devore Recognition Light System and Whelen Strobo. | | Phoenix Aviation Design Services Ltd. Change PADS/CP/1065. Master Document List PADS/MDL/1065/001 Rev. B. Flight Manual Supplements PADS/FMS/SA365/001, PADS/FMS/SA365/002, PADS/FMS/SA365/003. | AS365 NI, S/N 6309. | 27.08.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|-------------------|----------------|----------------------|---|--|--|---|--|---------------|
| 10017189 | PHOENIX AEROSPACE LIMITED | EASA.R.S.01326 | | | EUROCOPTER DEUTSCHLAND BK117 C | Installation of Honeywell AR30 Cockpit Voice Recorder | | Phoenix Aviation Design Services Modification PADS/CP/1067Phoenix Aviation Design Services AFM Supplement PADS/FMS/BK117/002 Revision B | -- | 11.01.2007 |
| 10017245 | PHOENIX AEROSPACE LIMITED | EASA.R.S.01442 | | | | Installation of Power Line Detector System. | | Master Document List PADS/MDL/1112/001RFM Supplement PADS/FMS/BK117/003. | Eurocopter Deutschland BK117 Series.RFM Supplement PADS/FMS/BK117/003 | 05.03.2008 |
| 10029133 | PHOENIX AEROSPACE LIMITED | P-EASA.R.S.01515 | | | AS355E, AS355F, AS355F1, AS355F2, AS355N, AS355NP | Installation of Skywatch L3 Traffic avoidance and Honeywell's KMD250 MFD | As specified on EASA TCDS R.146The Certification Basis for the original product remains applicable to this certificate/ approvalCS-27 change 2 dated 17 November 2008 used for this STCThe certified noise and/ or emissions levels of the original pro (...) | -PADS/FMS/AS355/001 revision A - Flight Manual Supplement or later revisions of the above listed documents approved by EASA-PADS/MDL/1115/001 revision A dated 26 February 2008 - Master Document List;- PADS/CP/1115 revision B dated 26 February 2008 - Chan (...) | Not Applicable | 08.03.2010 |
| 10041858 | PHOENIX AEROSPACE LIMITED | | | | A139 | Introduction of SX16 Searchlight System | CS-29 Initial IssueThe Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.The following paragraphs of CS29 Initial issue29.1, 29.25(>,<)> 29. (...) | Flight Manual Supplement; PADS/FMS/139/002 Rev. A or later revisions of the above listed documents approved by EASAMaster Drawing List; PADS/MDL/1197 Rev. BICA; PADS/ICA/1197 Rev A & PADS/MMS/1197/001 Rev A. | none | 11.12.2012 |
| 10041863 | PHOENIX AEROSPACE LIMITED | | | | AW 139 | Introduction of an electronic message sign. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.The following paragraphs of CS29 Initial issue29.1, 29.25, 29.45, 29.49, 29.51, 29.53. (...) | Flight manual supplement; PADS/FMS/139/001 Rev. A or later revisions of the above listed documents approved by EASAMaster drwg. List; PADS/MDL/1193 Rev. CICA; PADS/ICA/1193 Rev. A & PADS/MMS/1193/001 Rev. B | None. | 11.12.2012 |
| 10030088 | PIAGGIO AMERICA, Inc. | | | FAA STC SA10955SC | P.180 AVANTI & AVANTI II | FAA STC SA10955SC "Installation of a passenger seating system to replace five existing Deorane single seats with Model FS1101 single seats" | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | W & B Manual update Doc. AN-1802-WB-01 Rev. I (Instructions for Continued Airworthiness Doc. AN-1802-ICA-01 Rev. I) or later revisions of the above listed documents approved by EASA or later revisions of the above listed documents approved by EASA inacc (...) | None | 20.05.2010 |
| 10042554 | PIPER AIRCRAFT INC. | EASA.IM.A.C.01293 | | | Data not found | Installation of the Avdyne Entegra System Option (PFD/MFD) into the PA-34-Z20T (Seneca V). | | See EASA.IM.A.090 (TCDS) | See EASA.IM.A.090 | 26.06.2006 |
| 10016399 | PLANE POWER, LTD. | EASA.IM.A.S.02728 | | FAA STC SA10682SC | | Replacement of generator and mechanical voltage regulator with alternator and electronic voltage regulator | | FAA approved Master Drawing List, Doc. No. 14-0000, Rev. B, dated 31-Jul-2007, or later approved revision FAA approved Airplane Flight Manual Supplement, Doc. No. AFMS 001, Original Release, dated 25-Jul-2007, or later approved revision for Conversion (...) | The approval holder shall fulfil the obligations of Part 21, Paragraph 21.A109 | 13.05.2008 |
| 10040602 | PMV ENGINEERING | | REV. 1 | | A330-200, A330-300 | Installation of provisions for the use of EFB Class 2 on Airbus A330. This modification deals with the installation of actual Class 2 Electronic Flight Bag (EFB) CMC Esterline "Pilot View" system on Airbus A330-200/300. The modification is defined as follows (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are (...) | -Certification Investigation Report AVI-0034-CIR-R04 Issue 6, dated 12/07/2012- Instructions for Continued Airworthiness AVI-0034-ICA-R02 Issue 2, dated 25/06/2012- Aircraft Flight Manual Supplement AFMS AVI-0034-AFMS-R02 Issue 2, dated 09/03/2012- Aircra (...) | This certificate does not cover the installation of EDUs which will be covered by the operational evaluation. | 13.07.2012 |
| 10042279 | PMV ENGINEERING | | | | B747-400 | Installation of provisions for the use of EFB Class 2. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. EASA CS 25 Amendment 4/JAA TGL 36 - "Approval of Electronics Flight Bag (EFB)" The requ (...) | AVI-0035-CIR-R03AVI-0035-AFMS-R01AVI-0035-ICA-R01AVI-0035-MELS-R00AVI-0035-AMMS-R00 or later revisions of the above listed documents approved by EASA. | EASA STC 10040901 is a prerequisite for this modification. | 20.11.2012 |
| 10043587 | PMV ENGINEERING | | | | A330-201, A330-202, A330-203, A330-223, A330-243, A330-301, A330-302, A330-303, A330-321, A330-322, A330-323, A330-341, A330-342, A330-343. | Installation of ECS EZMount mechanical mounting on Airbus A330. | The Certification Basis for the original product remains applicable to this certificate/ approval. | Certification Investigation Report AVI-00250-CIR, Revision 0, dated 24/09/2012. Instructions for Continued Airworthiness AVI-00250-ICA, Revision 0, dated 24/09/2012. Aircraft Flight Manual Supplement AVI-00250-AFMS, Revision 0, dated 24/09/2012 or later (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 08.02.2013 |
| 10046097 | PMV ENGINEERING | | | | B737-600/700/800/900/900ER | Boeing-CMC Class 3 EFB system Operating System installation on B737NG. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification Investigation Report AVI-00450-CIR Rev.00. Instructions for continued airworthiness AVI-00450-ICA Rev.00; or later revisions of the above listed documents approved by EASA. | The installation of an EFB according to certificate 10044973 or 10045157 is a prerequisite for the installation of the above-mentioned Operating System. Prior to installation of this design change it must be determined that the interrelationship between (...) | 14.08.2013 |
| 10046214 | PMV ENGINEERING | | | | A330-201 / -202 / -203 A330-223 / -243 A330-301 / -302 / -303 A330-321 / -322 / -323 A330-341 / -342 / -343 | In-Flight Entertainment System Installation on A330. This STC installs IFE System, developed by the IMS Company. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification Investigation Report, ref : SCA-00210-CIR Revision 5, dated 29 August 2013; Instruction for Continued Airworthiness, ref : SCA-00210-ICA, dated 16 May 2013; or later revisions of the above listed documents approved by EASA. | 1-, The IFE system installation, defined in this STC, is only valid for a seat configuration identical to the one defined in PMV eng. Document ref : SCA-00210-CIR Revision 5, dated 29 August 2013.2-, This STC does not constitute the approval of the sea (...) | 30.08.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|--------------------|----------------|----------------------|--|---|--|---|---|---------------|
| 10017116 | POLICE AVIATION SERVICES Ltd. | EASA.R.S.01098 | | | BELL 412 EP | Installation of an L-3 Modular Airborne Data Recording/Acquisition System (MADRAS) combined CVR & FDR | | PAS Rotorcraft Flight Manual Supplement FMS/465-008 | | 03.05.2006 |
| 10017204 | POLICE AVIATION SERVICES Ltd. | EASA.R.S.01353 | | | MD HELICOPTERS MD902 | MTWA Increase with PAS External Stores and Surveillance Equipment | | FMS/496-002. | Only those in the above F/M Supplement. | 03.04.2007 |
| 10033979 | POLICE AVIATION SERVICES Ltd. | EASA STC 2004-3846 | REV. 1 | | EC 135 P1 (CDS) EC 135 P1 (CPDS) EC 135 P2 (CPDS) EC 135 P2+ EC 135 T1 (CDS) EC 135 T1 (CPDS) EC 135 T2 (CPDS) EC 135 T2+ EC 635 P2+ EC 635 T1 (CPDS) EC 635 T2+ | Incubator Installation EC 135 Transfer of EASA STC No 2004 3846 (dated 16/04/2004) from PAS Ltd to Bond Aviation Group Ltd | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate. | CAA 21NE2.00089; EASA Approval number 2004-03846; dated 16 April 2004 (for PAS modification PAS/MOD/388) | | 28.02.2010 |
| 10034324 | POLICE AVIATION SERVICES Ltd. | | | | MD902 CONFIG. P&W 206&207 | Aircraft Role Configuration | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-27 Amdt. 2 The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. (...) | RFM Supplement; FMS/592-002 Rev A or later revisions of the above listed documents approved by EASA Build standard; PAS/592-100-07 Compliance Report; DR/592-001 Rev C.ICA; MSS/592-008 Rev A. | none | 25.03.2011 |
| 10036834 | POLICE AVIATION SERVICES Ltd. | | REV. 1 | | VULCANAIR P.68.TC 'OBSERVER' | Airborne Surveillance System Installation | CS-23 Amendment 2 dated 28 September 2010 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. (...) | Modification Leaflet and Certificate of Design PAS/MOD/643 Compliance Checklist CCL/643-003 Issue B Maintenance Schedule Supplement MSS/643-007 Issue A Aircraft Flight Manual NOR.10.707-4B Supplement FMS/643.002 Issue B or later revisions of the above listed documents approved by EASA | None | 24.11.2011 |
| 10044695 | POLICE AVIATION SERVICES Ltd. | | | | MD 902 WITH PW 260E & 207E | MD902 NVIS | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-27 Amdt. 2; 1, 301, 303, 305, 307, 321, 337, 341, 561, 601, 603, 605, 6 (...) | Flight Manual Supplement: FMS/669-002 Issue B or later revisions of the above listed documents approved by EASA. Design Report: DR/669-001 Issue A Compliance checklist: CCL/669-003 Issue A.ICA: MPS/669-008 Issue A | None | 30.04.2013 |
| 10046149 | POLICE AVIATION SERVICES Ltd. | | | | MD902 EXPLORER | MD902 NVIS (PAS/MOD/680) Modification of aircraft to enable Night Vision Imaging System (NVIS) certification using Class B Night Vision Goggles | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual FMS/680-002 Issue A dated 13/08/2013 or later revisions of the above listed documents approved by EASA - Post Installation Actions, document PIA/680-007 Issue B - Maintenance Programme Supplement MPS/680-008 Issue A - Drawing List DL/680-099 Issue A | Required Installation of EASA STC 10028376, Aerodynamic NVIS on MD900 Installation of NVIS/NVG compatible lighting does not include or imply approval for flight operations with Night Vision Goggles (NVG). The operator must receive approval for the Civi (...) | 22.08.2013 |
| 10046466 | POLICE AVIATION SERVICES Ltd. | | | | MD900 | Internal Cargo Restraint System. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: (By Elect to Comply); CS-27 (Amdt. 3); 1, 301, 303, 305, 307, 321, 337, 341, 521, 561, 573, 601, 603, 605, 609, 610, 611, 6 (...) | Flight Manual Supplement: FMS/629-003 Iss. A or later revisions of the above listed documents approved by EASA. Design Report: DR/629-001 Iss. A. Design Definition/Drawing list; PAS/629/099 Iss. A.ICA; MSS/629-004 Iss. A | 902 configuration with PW 206E & 207E. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 20.09.2013 |
| 10015338 | PRATT & WHITNEY | EASA.I.M.A.S.00373 | REV. 2 | | AIR TRACTOR AT-402 A AIR TRACTOR AT-501 AIR TRACTOR AT-502 B AIR TRACTOR AT-503 AIR TRACTOR AT-802 AIR TRACTOR AT-802 A BEECH 1900 BEECH 200 BEECH 300 BEECH C90 CESSNA 208 CESSNA 208A CESSNA 208B PILATUS PC-6/B-H2 PILATUS PC-6/B1-H2 PILATUS PC-6/B2-H2 PILATUS PC-6/B2-H4 | Installation of Pratt & Whitney Data Acquisition System ADAS+, p/n ADAS-K-010. | | Part 23 Approved Model List ADAS-T225-1 dated 19 March 2004 Master Document List ADAS-T-100-1 dated 28 March 2005 Altair Avionics Corporation Aircraft Flight Manual Document No. ADAS-T-250-1, dated 3 July 2003, Revision C dated 31 March 2005, or later approved revision | 1. Compatibility of this modification with other previously approved modifications must be determined by the installer. 2. This Major Change to STC EASA.I.M.A.S.00373 is issued as a result of the validation of the Altair Avionics Corporation STC (...) | 07.03.2008 |
| 10015554 | PRATT & WHITNEY | EASA.I.M.A.S.01451 | REV. 1 | FAA STC SA00196BO | BEECH 1900 BEECH 200 BEECH 300 BEECH C90 CESSNA 208 CESSNA 208A CESSNA 208B | Installation of Altair Avionics Corporation Data Transmission Unit in accordance with instructions listed in Approved Model List (AML) DTU-T-225-1, Rev - or later approved revisions. FAA STC SA00196BO refers | | Approved Model List (AML) DTU-T-225-1 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.04.2008 |
| 10015834 | PRATT & WHITNEY | EASA.I.M.A.S.01925 | REV 1 | FAA STC ST00228BO | DC-9-81 (MD-81) DC-9-82 (MD-82) DC-9-83 (MD-83) DC-9-87 (MD-87) MD-88 | Installation of P&W JT8D-200 Series Engines incorporating the improved forward Fan Case as described in the P&W SB JT8D6478, Rev.1, dated 1 August 2007. | | P&W Service Bulletin JT8D 6478, Rev.1, dated 1 August 2007. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Prior to installation of this modification the installer must determine that the interrelationship between this (...) | 04.10.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------|-------------------|----------------|--------------------------------|--|---|---------------------|--|---|---------------|
| 10016426 | PRATT & WHITNEY | EASA.IM.A.S.02765 | | FAA STC SA0009580, LBA SA 1346 | BEECH 1900 BEECH 1900C, AIRLINER BEECH 1900D BEECH 200 BEECH 200C BEECH 200CT BEECH 200T BEECH 300 BEECH 300LW BEECH A200 BEECH A200C BEECH A200CT BEECH B200 BEECH B200C BEECH B200CT BEECH B200T BEECH B300 BEECH B300C | Installation of Pratt & Whitney Engine Services, Inc. Twin Engine Aircraft Data Acquisition System ADAS/ADAS+ Upgrade, P/N TWIN-K-010 | | Pratt & Whitney Engine Services, Inc. Installation Manual, Document No. TWIN-I-010-1, Revision J, dated November 8, 2002, or later approved revision | 1. Pratt & Whitney Engine Services, Inc. Aircraft Flight Manual Supplement for Raytheon Aircraft Company Models 200, 200C, 200CT, 200T, A200, A200C, A200CT, B200, B200C, B200CT, B200T, 300, 300LW, B300C, 1900, 1900C, 1900D, Document No. TWIN-T-250-1, (...) | 07.03.2008 |
| 10016692 | PRATT & WHITNEY | EASA.IM.E.S.01006 | | FAA STC SE00033EN | CFM56-3B1 | Installation of Pratt & Whitney (P&W) Global Material Solutions (GMS) CFM56-3 High Pressure Compressor Module in accordance with P&W GMS Drawing List 880500-01C dated May 06, 2008 or later EASA-approved revision EASA Certification Basis:1. Airworthiness Standards:JAR-E Change 2 (BCAR Issue 9 dated | | Refer to P&W GMS Service Bulletin PWGMS 72-2 Revision 2, dated December 17, 2008, and later approved revision. | Limitations:1. The limitations of the EASA Engine Type Certificate Data Sheet E.066 issue 1, dated November 28, 2008 apply to this Supplemental Type Certificate.2. Life limits for the parts approved by this Supplemental Type Certificate are pub (...) | 22.12.2008 |
| 10016693 | PRATT & WHITNEY | EASA.IM.E.S.01007 | | FAA STC SE00034EN | CFM56-3B1 | Installation of Pratt & Whitney (P&W) Global Material Solutions (GMS) CFM56-3 High Pressure Turbine Module in accordance with P&W GMS Drawing List 880500-01D dated May 06, 2008 or later EASA-approved revision EASA Certification Basis:1. Airworthiness (...) | | Refer to P&W GMS Service Bulletin PWGMS 72-3 revision 2, dated December 17, 2008, and later approved revision. | Limitations:1. The limitations of the EASA Engine Type Certificate Data Sheet E.066 issue 1, dated November 28, 2008 apply to this Supplemental Type Certificate.2. Life limits for the parts approved by this Supplemental Type Certificate are pub (...) | 22.12.2008 |
| 10016694 | PRATT & WHITNEY | EASA.IM.E.S.01008 | | FAA STC SE00035EN | CFM56-3B1 | Installation of Pratt & Whitney (P&W) Global Material Solutions (GMS) CFM56-3 Low Pressure Turbine Module in accordance with P&W GMS Drawing List 880500-01E dated May 06, 2008 or later EASA-approved revision EASA Certification Basis:1. Airworthiness (...) | | Refer to P&W GMS Service Bulletin PWGMS 72-4 revision 2, dated December 17, 2008, and later approved revision. | Limitations:1. The limitations of the EASA Engine Type Certificate Data Sheet E.066 issue 1, dated November 28, 2008 apply to this Supplemental Type Certificate.2. Life limits for the parts approved by this Supplemental Type Certificate are pub (...) | 22.12.2008 |
| 10016695 | PRATT & WHITNEY | EASA.IM.E.S.01009 | | FAA STC SE00032EN | CFM56-3B1 | Installation of Pratt & Whitney (P&W) Global Material Solutions (GMS) CFM56-3 Fan and Booster Module in accordance with P&W GMS Drawing List 880500-01B dated March 14, 2008 or later EASA-approved revision EASA Certification Basis:1. Airworthiness S (...) | | Refer to P&W GMS Service Bulletin PWGMS 72-1 revision 2, dated December 17, 2008, and later approved revision. | Limitations:1. The limitations of the EASA Engine Type Certificate Data Sheet E.066 issue 1, dated November 28, 2008 apply to this Supplemental Type Certificate.2. Life limits for the parts approved by this Supplemental Type Certificate are pub (...) | 22.12.2008 |
| 10016702 | PRATT & WHITNEY | EASA.IM.E.S.01017 | | FAA STC SE00026EN | ROLLS-ROYCE 250-C18 ROLLS-ROYCE 250-C18A ROLLS-ROYCE 250-C20 ROLLS-ROYCE 250-C20B ROLLS-ROYCE 250-C20F ROLLS-ROYCE 250-C20J ROLLS-ROYCE 250-C20R/1 ROLLS-ROYCE 250-C20R/2 ROLLS-ROYCE 250-C20S | Installation of Pratt & Whitney Engine Services, Inc. IntelliStart Plus +TM Electronic Control Unit (ECU), P/N DPU-K-010 on the specified 250 engine models. | | Airframe installation of the Altair Avionics Corporation - AlliedSignal Aerospace IntelliStart Plus +TM to be accomplished in accordance with applicable airframe STC. | 1. The IntelliStart Plus +TM ECU is designed to limit start temperatures and to provide an electronic aid for recording engine trend and cycle data.2. The IntelliStart Plus +TM ECU is designed as non-essential, ground use only equipment and cannot be (...) | 14.03.2008 |
| 10016703 | PRATT & WHITNEY | EASA.IM.E.S.01018 | | FAA STC SE00029EN, LBA EN0034 | HONEYWELL INT. T5313B HONEYWELL INT. T5317 | Installation of Pratt & Whitney Engine Services, Inc. SmartCycle Plus Engine Monitoring System, P/N EMU-K-010-1 on the specified T53 engine models | | | 1. SmartCycle Plus Engine Monitoring System is designed as an electronic aid for recording engine trend and cycle data. This system does not eliminate any scheduled maintenance or parts replacement based on time or any engine Instructions for Continue (...) | 04.03.2008 |
| 10016704 | PRATT & WHITNEY | EASA.IM.E.S.01019 | | FAA STC SE00027EN | ROLLS RROYCE 250-C | Installation of Pratt & Whitney Engine Services, Inc. IntelliStart Plus +TM Electronic Control Unit (ECU), P/N DPU-K-010 on the specified engine models | | Airframe installation of the Pratt & Whitney Engine Services, Inc. IntelliStart Plus +TM to be accomplished in accordance with applicable airframe STCs. | 1. The IntelliStart Plus +TM ECU is designed to limit start temperatures and to provide an electronic aid for recording engine trend and cycle data.2. The IntelliStart Plus +TM ECU is designed as non-essential, ground use only equipment and cannot be (...) | 14.03.2008 |
| 10016707 | PRATT & WHITNEY | EASA.IM.R.S.0048 | REV. 1 | N 2004-6170 | AGUSTA AB-412 AGUSTA AB-412EP BELL 212 BELL 222 BELL 222B BELL 222U BELL 412 BELL 412 EP | Installation of Pratt & Whitney Engine Services, Inc. SmartCycle monitoring and recording system | | Installation Manual Instruction for Continued Airworthiness as listed in document ref. AML SR00170B0, or later approved revision | 1. Required maintenance, as prescribed in the applicable airframe or engine Instructions for Continued Airworthiness, shall not be predicted on the data obtained from Pratt & Whitney Engine Services, Inc., monitoring/recording system.2. This system (...) | 25.03.2008 |
| 10016799 | PRATT & WHITNEY | EASA.IM.R.S.01157 | REV. 1 | | AGUSTA AB-412 AGUSTA AB-412EP BELL 212 BELL 412 BELL 412 EP | Installation of Pratt & Whitney Engine Services, Inc. Data Transmission Unit | | Instructions for Continued Airworthiness (ICA) as listed in document ref. AML DTU-T-225-4 Rev. A, dated September 29th, 2005 or later approved revision. | | 07.03.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------|-------------------|----------------|--------------------------------|--|---|---------------------|---|---|---------------|
| 10016803 | PRATT & WHITNEY | EASA.IM.R.S.01163 | REV. 1 | | SCHWEIZER AIRCRAFT 269D | Installation of Pratt & Whitney Engine Services, Inc. IntelliStart+ Electronic Control System and Gine Trend Monitor P/N DPU-K-010-14 | | Pratt & Whitney Engine Services, Inc. IntelliStart+ Installation Instructions DPU-I-010-6, Revision A, dated October 12, 2005, or later approved revisions. Pratt & Whitney Engine Services, Inc. Rotorcraft Flight Manual DPU-T-250-4, dated November 1, (...) | 1. Engine must be modified in accordance with STC SE00027EN, which must be EASA approved or accepted prior to installation. 2. Reference STC SE0027EN for any additional limitations applicable to the engine installation. 3. This system is designed as an (...) | 14.03.2008 |
| 10016827 | PRATT & WHITNEY | EASA.IM.R.S.01197 | REV. 1 | | AGUSTA A-109 E | Installation of a Pratt & Whitney Engine Services, Inc. Digital Aircraft Data Acquisition System ADASd | | Pratt & Whitney Engine Services, Inc. Aircraft Flight Manual Supplement, Document No. DAAS-T-250-3, Revision A, dated February 21, 2006, or later FAA-approved revision-Instruction for Continued Airworthiness (ICA) as listed in AML DAAS-T-225-1 Rev. A dated 21 February 2006 | 1. Required maintenance, as prescribed in the applicable airframe or engine Instructions for Continued Airworthiness, shall not be predicated on the data obtained from the Pratt & Whitney Engine Services, Inc. monitoring/recording system. 2. This system (...) | 11.03.2008 |
| 10016848 | PRATT & WHITNEY | EASA.IM.R.S.01224 | REV. 1 | | AGUSTA A-109 E | Installation of a Pratt & Whitney Engine Services, Inc. Data Transmission Unit Data Transmission Unit DTU-K-010-7 | | Instructions for Continued Airworthiness as listed in AML DTU-T-225-5, Revision, dated March 27, 2006, or later approved revision | 1. 109E with Pr & W: 206C Engine; | 07.03.2008 |
| 10017005 | PRATT & WHITNEY | EASA.IM.R.S.01417 | | FAA STC SR00125BO | MD HELICOPTERS 369A MD HELICOPTERS 369D MD HELICOPTERS 369E MD HELICOPTERS 369F MD HELICOPTERS 369FF MD HELICOPTERS 369H MD HELICOPTERS 369HE MD HELICOPTERS 369HM MD HELICOPTERS 369HS MD HELICOPTERS 500N | Installation of a Pratt & Whitney Engine Services, Inc. Honeywell IntelliStart Plus+TM Electronic Control System P/N DPU-K-010. Note: This STC reflects the transfer from Altair Avionics Corporation to Pratt & Whitney Engine Services, Inc. | | Pratt & Whitney Engine Services, Inc. Honeywell Installation Manual, DPU-I-010-3, Revision A, dated July 18, 2000, or later approved revisions. | 1. Aircraft must have engine modified in accordance with EASA STC EASA.IM.E.S.01019, FAA STC SE00027EN. 2. Pratt & Whitney Engine Services, Inc. Rotorcraft Flight Manual DPPU-T-250-2 dated July 27, 2000 or later approved revision, is required. 3. (...) | 14.03.2008 |
| 10017006 | PRATT & WHITNEY | EASA.IM.R.S.01418 | | FAA STC SR00113BO | BELL 206A BELL 206B BELL 206L BELL 206L-1 BELL 206L-3 BELL 206L-4 | Installation of a Pratt & Whitney Engine Services, Inc. AlliedSignal Aerospace IntelliStart Plus + TM Electronic Control System P/N DPU-K-010. Note: This STC reflects the transfer from Altair Avionics Corporation to Pratt & Whitney Engine Services, Inc. | | Pratt & Whitney Engine Services, Inc. AlliedSignal Aerospace Installation Manual, DPU-I-010-2, Revision E, dated January 14, 2000, or later approved revisions | 1. Aircraft must have engine modified in accordance with EASA STC EASA.IM.E.S.01017, FAA STC SE00026EN. 2. Compatibility of this design change with previously approved modifications must be determined by the installer. 3. Pratt & Whitney Engine (...) | 14.03.2008 |
| 10017007 | PRATT & WHITNEY | EASA.IM.R.S.01419 | REV. 1 | FAA STC SR00169BO, LBA RC1237 | EUROCOPTER DEUTSCHLAND BO-105 EUROCOPTER DEUTSCHLAND BO-105D EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 EUROCOPTER FRANCE AS 355N | Installation of Pratt & Whitney Engine Services, Inc. SmartCycle+ monitoring and recording system | | AML SR00169BO | 1. Required maintenance, as prescribed in the applicable airframe or engine Instructions for Continued Airworthiness, shall not be predicated on the data obtained from the Pratt & Whitney Engine Services, Inc. monitoring/recording system. 2. This sys (...) | 25.03.2008 |
| 10017008 | PRATT & WHITNEY | EASA.IM.R.S.01420 | | FAA STC SR00126BO AND LBA RC 1 | BELL 205A-1 | Installation of a Pratt & Whitney Engine Services, Inc. SmartCycle Plus System P/N EMU-K-010-2 | | Pratt & Whitney Engine Services, Inc. Installation Manual, EMU-I-010-2, Revision B, dated August 22, 2000 or later approved revisions | 1. Pratt & Whitney Engine Services, Inc. Rotorcraft Flight Manual EMU-T-250-1 dated August 31, 2000 or later approved revision, is required. 2. This system is designed as an electronic aid for recording engine trend and cycle data. This system does n (...) | 03.03.2008 |
| 10017009 | PRATT & WHITNEY | EASA.IM.R.S.01421 | | FAA STC SR00118BO, LBA STC RC | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA | Installation of Pratt & Whitney Engine Services, Inc. SmartCycle Engine Monitoring System, Part no. CYCLE-K-010 | | Pratt & Whitney Engine Services, Inc. Installation Manual, Doc. No. CYCLE-I-010-1, Revision C, dated March 29, 2000, or later approved revision | 1. Pratt & Whitney Engine Services, Inc. Aircraft Flight Manual Supplement for Eurocopter France, Models, AS-350B, AS-350B1, AS-350B2, AS-350BA, and AS-350B3, Doc. No. Cycle-T-250-1, Revision-, approved April 11, 2000, or later approved revision, (...) | 04.03.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|---------------------|----------------|--------------------------------|---|--|--|--|--|---------------|
| 10017010 | PRATT & WHITNEY | EASA.IM.R.S.01423 | | FAA STC SR000136BO, LBA RC1176 | EUROCOPTER FRANCE AS 355E EUROCOPTER FRANCE AS 355F EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 EUROCOPTER FRANCE AS 355N | Installation of a Pratt & Whitney Engine Services, Inc. SmartCycle Twin Engine Monitoring System Part No. CYCLE-K-010-4for AS355N and -5 for AS355E, AS355-F, AS355F1, AS355F2 | | Pratt & Whitney Engine Services, Inc. Installation Manual, Doc. No CYCLE-I-010-3, Revision D, dated November 30, 2000, or later approved revisions | 1. Pratt & Whitney Engine Services, Inc., Aircraft Flight Manual Supplement for Eurocopter France, Models AS355E, AS355-F, AS355-F1, AS355F2, and AS355N, Doc. No. CYCLE-T-250-2, Revision-, approved May 22, 2001, or later approved revision, is require (...) | 04.03.2008 |
| 10017011 | PRATT & WHITNEY | EASA.IM.R.S.01424 | | FAA STC ST00049BO, LBA RC 1048 | ROBINSON R22 ROBINSON R22 ALPHA ROBINSON R22 BETA | Installation of a Pratt & Whitney Engine Services, Inc. Health Usage Monitoring System (HUMS) Model 1022, P/N HUMS-A-030-1 | | HUMS Installation Instructions, HUMS-I-030-1, Version 1.3, dated December 8, 1995, or later approved revisions | 1. Federal Aviation Administration approval of the HUMS installation was determined on a non-interference basis. Operation of maintenance must not be predicated on the HUMS. | 04.03.2008 |
| 10017012 | PRATT & WHITNEY | EASA.IM.R.S.01425 | | FAA STC SR00067BO, LBA STC RC | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 BA | Installation of a Pratt & Whitney Engine Services, Inc. Health Usage Monitoring System (HUMS) Model 1122, P/N HUMS-A-070-1, Revision B | | HUMS Installation Instructions, HUMS-I-07003, HUMS-I-070-3, Version 1.4, dated November 13, 1996, or later approved revisions. | 1. Federal Aviation Administration approval of the HUMS installation was determined on a non-interference basis. Operation of maintenance must not be predicated on the HUMS. | 04.03.2008 |
| 10026953 | PRATT & WHITNEY | P-EASA.IM.A.S.03229 | | FAA STC ST00261BO | FALCON 2000EX | EASA Validation of FAA STC ST00261BO. Installation of Pratt <->&-> Whitney Engine Services, Inc Data Transmission Unit (DTU) in accordance with Pratt <->&-> Whitney Engine Services, Inc. Master Document List DTU-T-100-6, Revision F dated 23rd July 2000 (...) | On an elect to comply with basis, compliance has been shown with EASACS25 up to amendment 6 level, as per Pratt <->&-> Whitney Engine Services, Inc *EASA STC Compliance Report DTU-T-139-2 Revision B dated 16th July 2009 The certificated noise and/ or e (...) | Pratt <->&-> Whitney Engine Services, Inc Installation Instructions DTU-G-010-7, Revision C dated 26th May 2009- Pratt <->&-> Whitney Engine Services, Inc Instructions for Continued Airworthiness (ICA) DTU-G-260-7, Revision C dated 26th May 2009 or I (...) | Instructions for Continued Airworthiness (ICA) DTU-G-260-7, Revision C dated 26th May 2009 or later revisions shall be made available to the operator at the time of installation | 26.08.2009 |
| 10028632 | PRATT & WHITNEY | EASA.IM.A.S.03230 | | | GULFSTREAM 200 | | | | | 17.07.2009 |
| 10039147 | PRATT & WHITNEY ENGINE SERVICES, IN | | | FAA STC SA00303BO | C90GT, C90GT1 | ADAS for Hawker Beechcraft Model C90GT, C90GT1 Installation of the Pratt & Whitney Engine Services ADAS in accordance with Pratt & Whitney Engine Services# Installation Manual DAAS-G-010-6, Revision B, dated April 7, 2010, or later FAA approved revision. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. As established by the FAA STC SA00303BO. The requirements for environmental protection (...) | 1. 250-6 for Hawker Beechcraft Corporation Models C90GT and C90GT1 Part 23 aircraft, Revision A, approved July 1, 2010, or later FAA approved revision, is required with this modification. 2. Instructions for Continued Airworthiness (ICA), Pratt & Whitney (...) | None | 12.04.2012 |
| 10039150 | PRATT & WHITNEY ENGINE SERVICES, IN | | | FAA STC SA00295BO | 1. B200, B200C, B300, B300C 2. 300 | ADAS for Hawker Beechcraft Model B200, B200C, 300, B300, B300C Installation of the Pratt & Whitney Engine Services# Master Document List No. DAAS-T-100-3, Revision D, dated May 13, 2010, or later FAA-approved revision. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. As established by FAA STC SA00295BO. The requirements for environmental protection and (...) | 1. Pratt & Whitney Engine Services, Flight Manual Supplement No. DAAS-T-250-5 for Hawker Beechcraft Corporation Models B200, B200C, 300, B300, and B300C Part 23 aircraft, Revision D, approved May 14, 2010, or later approved revision, is required with t (...) | None | 12.04.2012 |
| 10043207 | PRATT & WHITNEY ENGINE SERVICES, IN | | | FAA STC ST11053SC | DHC-8-401 DHC-8-402 | Installation of a P & W Flight Data Acquisition, Storage and Transmission System (FAST). | CS 25 as per EASA.IM.A.191, Issue 6 (incl. EWS) The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 25, Amendment 11 The requirements for (...) | Installation: Must be in accordance with Pratt & Whitney Installation Data List GN4-T-100-1, Revision B Operation: Must be in accordance with Airplane Flight Manual Supplement GN4-T-250-1, Revision A. Maintenance: Must be in accordance with Instructions for (...) | None. | 11.01.2013 |
| 10016093 | PRECISE FLIGHT, Inc. | EASA.IM.A.S.02280 | REV. 2 | FAA STC SA01708SE | SR22, SR22T | Installation of Cirrus Fixed Oxygen System | The Certification Basis for the original product remains applicable to this certificate/ approval. CS 23, EASA certification basis defined in TCDS IM.A.007 ref CRI A-011 This certificate/ approval involves a change to the certificated noise and/ or emissions levels | Manufactured in accordance with Precise Flight Engineering Drawing Lists 102N0000-Rev. F, 100N0000-Rev. AH, 051A0000-Rev. B, 027N0000-Rev. E, 026N0000-Rev. E, 020N0000-Rev. G, 010N0000-Rev. H, 016N0000-Rev. C, 012N0000-Rev. IR, 011N0000-Rev. A, 010A0000-R (...) | - according limitation section of AFM-S- max altitude extension up to 25.000 ft, if operated acc. Limitation section of AFM-S | 03.08.2010 |
| 10016858 | PRECISE FLIGHT, Inc. | EASA.IM.R.S.01234 | REV. 1 | FAA STC SH331NM | SEE EASA APPROVED MODEL LIST | Precise Flight PulseLite Control System. Adding MDHI 369 and 500N models. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Multiple (see attached sheet) | Approval for the physical installation of the PulseLite Control Unit will be required if there is no specific reference provided in Engineering Drawing List 001P0001, revision B, dated May 1, 2002, or later FAA approved revision. | 08.04.2011 |
| 10016859 | PRECISE FLIGHT, Inc. | EASA.IM.R.S.01235 | | FAA STC SA00701SE | ROBINSON R22 & R44 | Installation of Precise Flight PulseLite System in accordance with Precise Flight, Inc. Engineering drawings 590P0001, Rev. -, dated 26 January, 1999 and 590P0002, Rev. -, dated 26 January 1999 or later approved revision in Robinson R22 model or R44 model helicopters. | | RFMS: Precise Flight, Inc. FAA approved Rotorcraft Flight Manual Supplement, dated May 18, 1999 or later approved revisions. Installation Instructions and Instructions for Continued Airworthiness *Installation Instructions No.: PPRI-3000* Revision 9 or later revision. | | 20.07.2007 |
| 10017060 | PRECISE FLIGHT, Inc. | EASA.IM.R.S.01494 | | FAA STC SR01821SE | R22 MARINER R22, R22 ALPHA, R22 BETA | Installation of a Precise Light HID Aviation Lighting System according to FAA STC SR01821SE. | | 1. Engineering Drawing List (Manufacturing) 020H0000 Rev. C2. Engineering Drawing List (Installation) 590H0000 Rev. B3. Instructions for Continued Airworthiness 590HMAN0002 Rev. A | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 14.11.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------|----------------|----------------------|---------------------------|--|---|---|---|---------------|
| 10035215 | PRECISE FLIGHT, Inc. | | | FAA STC SA2622NM | FALCON 50 | Installation of Precise Flight Pulse Light Control system in Falcon 50aircraft.EASA validation of FAA STC SA2622NM for Falcon 50 aircraft. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated nossand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Engineering Drawing List doc N° 001P0000 Rev G dated 08/08/2007.Aircraft Flight Manual Supplement listed on FAA AML N° SA2622NM dated 06December 2004.Installation Manual listed onFAA AML N° SA2622NM dated 06 December2004.Instructions for Continued A (...) | This approval is restricted to the following Pulselite units:2400-25-4-4 / 2400-25-4-4-1 / 2400-25-4-4-2 / 2401S / 2401.This approval does not cover the physical installation of the system.Physical installation of the system in the aircraft should b (...) | 06.06.2011 |
| 10015539 | PRECISION CONVERSIONS L.L.C. | EASA.IM.A.S.01423 | | ST1529SE | BOEING 757-200 SERIES | Conversion from Passenger to Freighter Configuration (including Main Deck Cargo Door, Crew Entry Door, Interior & Associated Systems. | | Precision Conversions Master Drawing List PC-0118-09 Revision M dated July 18 2006 or later approved revision.PrecisionConversions Service Bulletin 757-52-0006 Revision Originaldated October 11 2006.Precision Conversions AFM Supplement OLO181, OEMS (...) | | 27.10.2006 |
| 10014487 | PREMIAIR AIRCRAFT ENGINEERING | EASA.A.S.02549 | | | BEECH 390 | Satellite Telephone Installation | | Premiair Aircraft Engineering Modification 06-001 | | 11.05.2006 |
| 10017110 | PREMIAIR AIRCRAFT ENGINEERING | EASA.R.S.01087 | | | AS 365N2 | Premiair Aircraft Engineering Modification 05-010.KLN 900 GPS Receiver Installation. | | Premiair Aircraft Engineering Modification Summary Sheet 05-010.Premiair Aircraft Engineering AFM Supplement 05-AS365N2-01 Issue 1 dated July 01 2005 or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 10.08.2005 |
| 10017111 | PREMIAIR AIRCRAFT ENGINEERING | EASA.R.S.01088 | | | AS 365N2 | Premiair Aircraft Engineering Modifications 05-007 & 05-009. Modifications embodied, since manufacture, in Brazil & South Africa. Also modifications embodied to meet EASA & JAR OPS-3 requirements. | | Premiair Aircraft Engineering Modifications 05-007 & 05-009. Premiair Aircraft Engineering AFM Supplements 05AS365N2-04 Issue 1. 05-AS365N2-03 Issue 1 & 05-AS365N2-02 Issue 1 or later approved revisions. | 1. Aircraft Serial No. 6393 Only2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.08.2005 |
| 10017121 | PREMIAIR AIRCRAFT ENGINEERING | EASA.R.S.01208 | | | AS 355N | IFR Upgrade with EFIS 40 Installation | | 1. Modification Summary Sheet & Declaration of Compliance 04-024 EFIS 40 & Single Pilot IFR Certification2. AFM Supplements 04-AS355N-3 Issue 1, 04-AS355N-4 Issue 1 | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 16.11.2005 |
| 10017155 | PREMIAIR AIRCRAFT ENGINEERING | EASA.R.S.01261 | REV. 2 | | AS365N2, AS365N3, SA365N1 | Installation of Honeywell MK XXI EGPWSNote: This revision is to amend an existing EASA STC (RefP-EASA.R.S.01261 rev. 1)to add the AS 365N3 model | The Certification Basis for the original product stated in TCDS N.159remains applicable to this certificate/ approval | -Installation in accordance with Premiair Aircraft Engineering Modification reference 09-026 issue 2.-Premiair Aircraft Engineering RFM Supplement 09-AS365N3-03 Issue 1-Installation in accordance with Premiair Aircraft EngineeringModification refer (...) | None | 01.02.2010 |
| 10017160 | PREMIAIR AIRCRAFT ENGINEERING | EASA.R.S.01268 | | | AS 355 F1 | Install a Honeywell KLN90B GPS | | Premiair Aircraft Engineering Modification 06-014Premiair Aircraft Engineering RFM Supplement 06-AS355F1-01 Issue A or alter approved revision | -- | 07.07.2006 |
| 10017174 | PREMIAIR AIRCRAFT ENGINEERING | EASA.R.S.01289 | REV. 2 | | AS365N2, AS365N3, SA365N1 | Installation of Honeywell KTA 870 Traffic Advisory SystemNote: This revision is to amend an existing EASA STC (RefP-EASA.R.S.01289 rev. 1) to add the AS365N3 model | The Certification Basis for the original product stated in TCDS N.159remains applicable to this certificate/ approval | Installation in accordance with Premiair Aircraft EngineeringModification reference 09-025 issue 2.-Premiair Aircraft Engineering RFM Supplement 09-AS365N3-02 Issue 1-Installation in accordance with Premiair Aircraft EngineeringModification referen (...) | None | 01.02.2010 |
| 10017175 | PREMIAIR AIRCRAFT ENGINEERING | EASA.R.S.01290 | REV. 2 | | AS365N2, AS365N3, SA365N1 | Installation of Honeywell KMD 850 Multi Function DisplayNote: This revision is to amend an existing EASA STC (RefP-EASA.R.S.01290 rev. 1) to add the AS 365N3 model | The Certification Basis for the original product stated in TCDS N.159remains applicable to this certificate/ approval | -Installation in accordance with Premiair Aircraft EngineeringModification reference 09-024 issue 2.-Premiair Aircraft Engineering RFM Supplement 09-AS365N3-01 Issue 1-Installation in accordance with Premiair Aircraft EngineeringModification referen (...) | None | 01.02.2010 |
| 10017227 | PREMIAIR AIRCRAFT ENGINEERING | EASA.R.S.01397 | | | S-76B | Traffic Advisory System Installation. | | Premiair Aircraft Engineering Limited Modification 07-019Premiair Aircraft Engineering Limited Rotorcraft Flight Manual Supplement Nr: 07-S76B-04 Issue 1. | | 26.05.2008 |
| 10017233 | PREMIAIR AIRCRAFT ENGINEERING | EASA.R.S.01411 | | | AS 355 N | Installation of a Honeywell KMD850 Multifunction Display replacing the existing IN812A Radar Indicator plus GC360 RadarGraphics Computer and Interfacing with an existing Honeywell RDR2000. | | Engineering Modification 07-022.FMS : 07-AS355N-01. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 05.11.2007 |
| 10027205 | PREMIAIR AIRCRAFT ENGINEERING | P-EASA.A.S.03982 | | | BEECH 390 (PREMIER 1) | Cabin configuration change | CS 23 | Configuration change in accordance with Premiair modification No 09-015.Operation in accordance with Premiair Flight Manual Supplement No09-RA390-01 Issue 1, dated 24 August 2009,or later approved revisions. | None | 14.09.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|---------------------|----------------|----------------------|---|--|--|--|---|---------------|
| 10039783 | PREMIAIR AIRCRAFT ENGINEERING | | | | A-60+ | Installation of Cabin Safety Equipment | The Certification Basis for the original product remains applicable to this certificate/ approval. | Certification Program 12-007, Issue 3, 8 May 2012- Change/Repair amendment Form 12-007, Issue 2, 8 May 2012- Technical Report 12-007-002-002, Issue 2, 8 May 2012- Technical Report 12-007-002-006, Issue 1, 16 April 2012. | None | 23.05.2012 |
| 10015562 | PREMIER AIR CENTER, Inc. | EASA.IM.A.S.01460 | | ST01884CH | FAN JET FALCON SERIES D FAN JET FALCON SERIES E FAN JET FALCON SERIES F MYSTERE FALCON 20-C5 MYSTERE FALCON 20-D5 MYSTERE FALCON 20-E5 MYSTERE FALCON 20-F5 MYSTERE FALCON 200 | Installation of a Honeywell Mark VIII Enhanced Ground Proximity Warning System with Terrain Display | | See item 5 of approval report (attachment) | 1. EASA approved Flight Manual Supplement, Premier Air Center Document 11-327B001, Revision I/R, or later EASA approved revision is a required part of this STC (Configuration-01)2.EASA approved Flight Manual Supplement, Premier Air Center Document 11- (...) | 29.06.2006 |
| 10043311 | PREMIER AIR CENTER, Inc. | EASA.IM.A.S.02833 | | FAA STC ST02388CH | BD-100-1A10 (CHALLENGER 300) | Installation of a Rockwell Collins IFIS-5000 Integrated Flight Information System with display on the existing Rockwell Collins Proline21 Continuum Electronic Flight Instrument System (EFIS) / EASA Validation of FAA STC ST02388CH | | - Master Data List, ref. doc. 09-385L001, Rev. A- Airplane Flight Manual Supplements, ref. doc. 09-385B001, Rev. I/R & ref. doc. 09-385B003, Rev. I/R- Maintenance Requirements - Instructions for Continued Airworthiness, ref. doc. 09-385B002, Rev. A | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The (...) | 01.07.2009 |
| 10027600 | PREMIER AIRCRAFT, LLC | P-EASA.IM.A.S.02677 | | FAA STC ST01951LA | MF50 BASIC CONFIGURATION | EASA Validation of FAA STC ST01951LA Installation of TFE731-4-1C engine part number 3075900-1, per Honeywell Service Bulletin TFE731-72-3703, dated January 19, 2007 on the MF50 basic aircraft configuration in accordance with Flight Test Associates, Inc (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the certificated noise and/or emissions levels | Flight Test Associates, Inc. Master Drawing List (MDL) No. PA-00100, Revision A, dated 03rd October 2006- Aeroplane Flight Manual Supplement Document No. FTA-PA-001019 Revision C dated 30th September 2009- Flight Test Associates, Inc. Instructions f (...) | Not Applicable | 13.10.2009 |
| 10043216 | PRO STAR AVIATION, L.L.C. | | | FAA STC SA00321BO | PC-12/45, PC-12/47, PC-12/47E | STC Prostar Aviator 200 inst. on PC-12/45/47/47E Swift Broadband Communication with WiFi and Phone through SATCOM. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAA MoC IP S-1 stage 4 "Installation WLAN". The certificated noise and/or emissions l (...) | FAA AFMS "Aviator 200 Swift/broadband Sys" Doc. 21037-FMS-01 Rev. I/R for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | S/N: all legacy MSN | 14.01.2013 |
| 10029073 | PROTOPLANE SPECIAL MISSION AIRCRAFT | EASA.A.S.03952 | | | BEECH 200 | Video transmission mast under a Beech 200. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | PP-012-MDL-05, Master Data List, Issue 5, dated 1 March 2010 PP-012-AMM-03 Supplement to Aircraft Maintenance Manual, Rev.3 dated 5th February 2010 PP-012-ICA-00 Supplement to Instructions for Continued Airworthiness, Rev.0 dated 23rd of January 201 (...) | All associated Audio/Video Relay electronic systems and antennas must be turned off during VOR navigation and before beginning the approach. | 01.03.2010 |
| 10034751 | PROTOPLANE SPECIAL MISSION AIRCRAFT | | | | AS 350 B, -D, -B1, -B2 AS 350 BA, -BB, -B3 | Installation of video transmission equipment. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | PP-013-AFM-00 Flight Manual Supplement or later revisions of the above listed documents approved by EASAPP-013-CP-00 dated 15 April 2010 - Certification Plan; PP-013-IM-02 issue 2 dated 4 April 2011 - Installation Manual; PP-013-MDL-02 issue 2 dated 4 April 2011 - Master Data List. | None | 28.04.2011 |
| 10038293 | PROTOPLANE SPECIAL MISSION AIRCRAFT | | | | BN-2 | BEZU installation on a BN-2 rev2 du 27/01/2012. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | None. | None. | 03.02.2012 |
| 10038491 | PROTOPLANE SPECIAL MISSION AIRCRAFT | | | | AS355E, AS355F, AS355F1 AS355F2, AS355N, AS355NP | Video transmission equipment | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | PP-014-AFM-00 issue 0 dated 15 July 2011 - Flight Manual Supplement or later revisions of the above listed documents approved by EASAPP-014-CP-00 dated 15 July 2011 - Certification Plan; PP-014-DC-00 dated 15 July 2011 - Declaration of Compliance; PP-01 (...) | When antenna holder PP-10-A601 is installed VNE is limited to 120 kts; When rack PP-10-A421 is installed, carriage of passengers in front seats is prohibited; When racks PP-10-A401c or PP-A461 are installed, a maximum of 2 passengers in rear seats is permitted. | 21.02.2012 |
| 10039039 | PROTOPLANE SPECIAL MISSION AIRCRAFT | | | | EC135 T1, T2, T2+, P1, P2, P2+ | Installation of a Max-Viz EVS 1500 vision enhancement system. Installation of EVS 1500 system for Protoplane change project PP-015 as defined in PP-015-MDL-01. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement PP-015-AFM-02 Rev. 02; or later revisions of the above listed documents approved by EASA; PP-015-CP-03 Certification Plan Issue 3; PP-015-ICA-00 Instruction for Continued Airworthiness Issue 0; PP-015-WD-01 Wiring Dia (...) | Limited to EC135 equipped with CPDS and FCDS. | 02.04.2012 |
| 10039373 | PROTOPLANE SPECIAL MISSION AIRCRAFT | | | | MF50 | Attachment provisions on Falcon 50 S/N 011. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List N° PP-033-MDL-01 Issue 1 dated 20.04.2012 Instructions for Continued Airworthiness N° PP-033-ICA-00 Issue 0 dated 01.04.2012 or later revisions of the above listed documents approved by EASA. | Limited to S/N 011. | 30.04.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|---------------------|----------------|--------------------------------|---|---|--|---|--|---------------|
| 10040714 | PROTOPLANE SPECIAL MISSION AIRCRAFT | | | | B200C | Removing of the metal parts of a cargo pod and provisions for the installation of signal transmission antennas inside the pod. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List, PP-022-MDL-00, Issue 2, dated 23/07/2012 Aircraft Flight Manual Supplement, PP-022-AFM-02, Issue 2, dated 23/07/2012 or later revisions of the above listed documents approved by EASA | Limited to serial numbers BL 49 and BB 1473It is forbidden to use the modified pod to carry baggage or cargo. This modification does not include power to the antennas installed on the pod. The operation of the antennas is not included in this approval and must be approved by a separate change. | 23.07.2012 |
| 10043552 | PROTOPLANE SPECIAL MISSION AIRCRAFT | | REV. 1 | | B200, B200C, B200GT, B200CGT B300, B300C (KING AIR) | Installation of a radar and a turret on a B350ER | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels. | Master Document List: PP-026-MDL-05 Instructions for Continued Airworthiness: PP-026-ICA-01 Flight Manual Supplement: PP-026-AFM-06 or later revisions of the above listed documents approved by EASA | Limited to S/N FL708 Limitations according to Flight Manual Supplement PP-026-AFM-06. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change (...) | 12.02.2013 |
| 10015666 | RAISBECK ENGINEERING Inc. | EASA, IM.A.S. 01642 | REV 3 | ST01468SE | LEARJET 35 LEARJET 35A LEARJET 36 LEARJET 36A | Installation of drag reduction system (including outboard trailing edge extension, leading edge vortons & contoured flap). | | 1. Raisbeck Engineering Drawing Number 202L-5700 Revision C dated March 4 2005 or later approved revision. 2. Raisbeck Engineering AFM Supplement 204L-5780 Revision IR dated March 4 2005 or later approved revision. | | 21.08.2006 |
| 10016131 | RAISBECK ENGINEERING Inc. | EASA, IM.A.S. 02325 | | STC ST01724SE DATED JANUARY ST | LEARJET 31 | Installation of a drag reduction system (contoured flap). | | AFM supplement 06L-5780, revision IR (or later approved revisions). | | 22.11.2007 |
| 10016246 | RAISBECK ENGINEERING Inc. | EASA, IM.A.S. 02521 | REV. 2 | FAA STC SA3593NM | KING AIR 65-A90, B90, C90 KING AIR 65-C90A, C90GT, C90GT KING AIR 65-C90GTI, E90 | Installation of propeller system. | FAR 23 | -Installed of Hartzell/Raisbeck Quiet Turbofan Propeller in accordance with Raisbeck Engineering Drawing List 86-2500 Revision AM dated October 13, 2009 or later approved revisions. -Various Approved Flight Manual Supplements specific to airplane models (...) | 1. See attached Propellers and Propeller Limitations. 2. Hartzell HC-D4N-3C hubs with Hartzell D9290K blades in pairs. Diameter 83.0 in. (maximum); minimum allowable for repair: 92.0 in. No further reduction in diameter permitted. Pitch Settings: Low Pitch (...) | 06.09.2010 |
| 10016579 | RAISBECK ENGINEERING Inc. | EASA, IM.A.S. 02983 | | FAA STC SA2939NM | BEECH 65-90 BEECH 65-A90 BEECH B90 BEECH C90 BEECH C90A | Installation of nacelle overwing storage lockers and lower flap fairing. | | 1. Installation Manual 006-1007 Rev. B, dated December 12th, 2007. 2. Installation Manual 006-1008 Rev. A, dated October 4th, 2006. 3. Installation Manual 85-1007 Rev. 1, dated April 27, 2007. 4. Approved Flight Manual Supplement 85-101 Rev (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 05.08.2008 |
| 10016580 | RAISBECK ENGINEERING Inc. | EASA, IM.A.S. 02984 | | FAA STC ST3857NM | BEECH 200 BEECH 200C BEECH 200CT BEECH 200T BEECH A200 BEECH A200C BEECH A200CT BEECH B200 BEECH B200C BEECH B200CT BEECH B200T | Installation of nacelle overwing storage lockers and lower flap fairing. | | 1. Installed in accordance with Raisbeck installation manual Installation Manual 200-5421 Rev. D, dated November 5, 2007. 2. Approved Flight Manual Supplement 87-105 Rev. C dated December 2007. 3. Instructions for continued Airworthiness ref. (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 05.08.2008 |
| 10016581 | RAISBECK ENGINEERING Inc. | EASA, IM.A.S. 02985 | | FAA STC SA3831NM | BEECH 200 BEECH 200C BEECH 200CT BEECH A200 BEECH A200C BEECH A200CT BEECH B200 BEECH B200C BEECH B200CT BEECH B200T | Installation of improved leading edge consisting of inboard leading edges, deice boots, leading edge ducts, leading edge to nacelle fairing, and associated hardware. | | 1. Installed in accordance with Raisbeck installation manual Installation Manual R81-1022, Rev. F, dated January 30th 2008. 2. Instructions for Continued Airworthiness Ref. Raisbeck Maintenance Manual 85-120 Rev. G, dated January 2008. 3. IPC (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 05.08.2008 |
| 10016582 | RAISBECK ENGINEERING Inc. | EASA, IM.A.S. 02987 | REV. 2 | FAA STC SA4010NM | 1. 65-90, 65-A90, B90, E90 2. C90A, C90GT, C90GTI | Installation of dual aft body strakes on the 90 series King Air aircraft per Raisbeck Engineering Inc. STC SA4010 NM. | | 1. Installed in accordance with Raisbeck installation manual Installation Manual 87-2024 Rev. E, dated 12/19/07. 2. Instructions for Continued airworthiness ref. Raisbeck Maintenance Manual 85-120 Rev. G. Dated January 2008. 3. IPC 85-119 dated April 2004 (...) | none | 26.04.2011 |
| 10016583 | RAISBECK ENGINEERING Inc. | EASA, IM.A.S. 02988 | | FAA STC SA3366NM | BEECH 200 BEECH 200C BEECH 200CT BEECH A200 BEECH A200C BEECH A200CT BEECH B200 BEECH B200C BEECH B200CT BEECH B200T | Installation of Ram Air Recovery System. | | 1. Installed in accordance with Raisbeck installation manual Installation Manual R81-1021 Rev. 1, dated February 8, 2006. 2. Approved Flight Manual Supplement 99-117 Revision A, dated February 1, 2008. 3. Instruction for Continued Airworthiness (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 05.08.2008 |
| 10016599 | RAISBECK ENGINEERING Inc. | EASA, IM.A.S. 03012 | REV. 1 | FAA STC ST3591NM | 200, 200C, 200T, 200CT, A200, A200C, A200CT, B200, B200C, B200GT, B200CGT, B200T, B200T, B200CT, | Installation of lower aft fuselage strakes. | | 1. Installed in accordance with Raisbeck installation manual Installation Manual 200-5421 Rev. D, dated November 5, 2007 or later approved revisions. 2. Approved Flight Manual Supplement 85-111 Revision C dated December 2007 or later approved revision. 3. (...) | None | 01.07.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|-------------------|----------------|----------------------|--|--|---|--|--|---------------|
| 10016620 | RAISBECK ENGINEERING Inc. | EASA.IM.A.S.03051 | | FAA STC ST00752SE | BEECH 200 SEE ATTACHED EASA APPROVED TYP | This change applies only to the above listed model aircraft in accordance with Raisbeck Engineering Master Drawing List99-1500, Revision D, dated May 13, 2008 or later approved revisions. | | Operation in accordance with FAA Approved Flight Manual Supplement 99-126, Revision A, dated January 10, 2000, 99-126H, dated October 4, 2007, or 08-B200GT-126, Revision R, dated May 13, 2008, or later approved revisions. | Requires the prior incorporation of Raisbeck STCs# ref.-SA4175NM (when equipped with High Floatation gear)/SA3366/NMSA3931/NMSA3519NMI is the installers responsibility to ensure all these referenced STCs are EASA approved. | 15.12.2008 |
| 10030124 | RAISBECK ENGINEERING Inc. | EASA.IM.A.S.02986 | REV. 3 | FAA STC SA4175NM | 200, 200C, 200T, 200CT, A200 A200C, A200CT, B200, B200C B200T, B200CT, B200GT, B200CGT B300, B300C | Installation of fully enclosed main landing gear door system for high-floatation gear equipped aircraft. | FAR 23. | Installed in accordance with Raisbeck Installation Manual R81-1005 Rev.N, dated 10 March 2010, Instructions for Continued Airworthiness ref. Raisbeck Maintenance Manual 85-120 Rev. G. Dated January 2008/IPC 85-119 dated April 2004 or later revisions of (...) | None. | 10.08.2011 |
| 10031275 | RAISBECK ENGINEERING Inc. | EASA.IM.A.S.00765 | REV. 2 | FAA STC SA2698NM-S | SEE EASA APPROVED MODEL LIST | Installation of Harzell/Raisbeck Quiet Turbofan or Swept Turbofan/Propellers. The installation consists of Harzell/Raisbeck 4-bladed propellers and aluminium hubs with spinners, and a propeller de-icing system. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Raisbeck Engineering Drawing List85-1500, Rev. AB, dated October 1, 2009; Operation of aircraft with Harzell D9383K blades installed to be in accordance with the AFM/POH supplements defined in the attached Table A. Oper (...) | Data is suitable for reproduction and installation on all Beech SuperKing Air Model 200 series aircraft. Approval of this change in type design applies to the above model aircraft only (further defined in Table A and B). This approval should not be ext (...) | 23.07.2013 |
| 10031307 | RAISBECK ENGINEERING Inc. | | | FAA STC ST01182LA | A320-232 | A320 bin extension system installation of a bin extension system in accordance with FAA Supplemental Type Certificate ST01182LA issued 10 July 2001. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Raisbeck Commercial Air Group (RCAG) Project Specific Certification Plan number 200A-2540 revision C dated 5 January 2001. - A320 Bin Extension Installation Part List 200A-2500 Rev. E dated 11 March 2002. - Master Data List 200A-2010 Rev. G dat (...) | Refer to paragraph "Limitations & Conditions" listed in the FAA STC ST01182LA. The Maintenance Instructions for Continued Airworthiness are contained in the RCAG Component Maintenance Manual 200A-2560. These maintenance instructions must be included i (...) | 09.08.2010 |
| 10027805 | RAM AIRCRAFT, LIMITED PARTNERSHIP | | | FAA STC SE10746SC-D | IO-520-B -, -BB, -C, -CB | Conversion of Teledyne Continental Motors (TCM) models IO-520-B, -BB to RAM model IO-550-B-RA engine, or conversion of TCM models IO-520-C, -CB to RAM model IO-550-C-RA engine in accordance with RAM Master Drawing List 2691, Revision I.R. dated November 7, 2006, or later FAA approved revision. | The Certification Basis for the original product remains applicable to this certificate/ approval. | Operator's Manual Supplement for RAM IO-550 series engines; RAM Document 2729, Rev. A, dated August 29, 2007 or later approved revision; Instructions for Continued Airworthiness: RAM Document 2692, dated November 7, 2006, or later approved revision; Overhaul (...) | The RAM model IO-550-B-RA engine is identical to and has the same ratings, limitations, and installation interfaces as TCM model IO-550-B engine. The RAM model IO-550-C-RA is identical to and has the same ratings, limitations, and installation interfaces as TCM model IO-550-C engine. | 17.11.2009 |
| 10031425 | RAM AIRCRAFT, LIMITED PARTNERSHIP | | REV. 2 | FAA STC SA8125SW-D | CESSNA 414A | Installation of vortex generators in acc. with FAA STC SA8125SW-D. Issued, dated May 12, 2008. Option 1: Modification in acc. with RAM Master Drawing No. 1507, Revision C, dated 11/03/1992. Option 2: Modification in acc. with RAM Master Drawing No. 1 (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certificated noise and/or emissions levels. | Option 1: No change to original Airplane Flight Manual. Option 2: (a) ,reserved(b) ,reserved(c) ,reserved(d) , RAM AFMS 1027, dated 16/03/1992(e) ,reserved(f) , RAM AFMS 1029, dated 16/03/1992 or later revisions of the above listed documents approved by E (...) | None | 12.11.2010 |
| 10031513 | RAM AIRCRAFT, LIMITED PARTNERSHIP | | REV. 1 | FAA STC SA09105SC-D | CESSNA 414A | Modification in acc. with FAA STC SA09105SC-D. Aircraft modified in accordance with RAM Master Drawing No. 1765, Revision B, dated 20/09/1995 and Installation of modified TCM engine model TSIO-520-NB. Option 1: Cessna 414A modified by FAA STC SE09104SC-D (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certificated noise and/or emissions levels. | Option 1: RAM AFMS 1080, dated 25/09/1995. Option 2: RAM AFMS 1082, dated 25/09/1995. Option 3: reserved. Option 4: RAM AFMS 1088, dated 25/09/1995 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 200 (...) | This approval is limited to Hartzell PHC-C3YF-2UF/FC7663D(B)-2Q or PHC-C3YF-2UF/FC7693DF(B) propellers. | 12.11.2010 |
| 10032348 | RAM AIRCRAFT, LIMITED PARTNERSHIP | | REV. 1 | FAA STC SA09971SC-D | 1. CESSNA 340, 340A 2. CESSNA 414, 414A, 402C | Installation of Hartzell propeller PHC-C3YF-2UF/FC7693DF(B) in accordance with FAA STC SA09971SC-D | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certificated noise and/or emissions levels. | RAM Drawing No. 2549, Revision D, dated May 24, 2001. Airplane Flight Manual Supplement AFMS 1096, Revision A, dated December 2, 2002. The Instruction of Continued Airworthiness are contained in Hartzell Propeller's Overhaul and Maintenance Manual No. 117 (...) | 1. Hartzell Model PHC-C3YF-2UF/FC7693DF or PHC-C3YF-2UF/FC7693DF propellers are limited to aircraft that are previously FAA approved to install 78-inch (198 cm) diameter propellers and have TCM model TSIO-520 series engines with maximum rated power f (...) | 02.11.2010 |
| 10038286 | RAM AIRCRAFT, LIMITED PARTNERSHIP | | | | CESSNA 414 | Modification of Cessna 414 in acc. with FAA STC SA09469SC-D. Aircraft modified in accordance with RAM Master Drawing No. 1982, Revision K, FAA Approved September 23, 2009 and Installation of modified TCM engine model TSIO-520-NB in accordance with FAA STC SE09261SC-D or SE09104SC-D. | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certificated noise and/or emissions levels. | RAM AFMS 1087, Revision C, dated 25/09/1995, FAA approved September 08, 2008. Installation Instruction: Master Drawing 1982, Rev. K, FAA approved September 23, 2009 or later revisions of the above listed documents approved by EASA in accordance with EAS (...) | This approval is limited to Hartzell propellers: PHC-C3YF-2UF/FC7663D(B)-2Q | 03.02.2012 |
| 10015723 | RAYTHEON AIRCRAFT COMPANY | EASA.IM.A.S.01731 | | FAA STC SA8351SW | BAE 125-1000A | Installation of Brackets for Cabin Headliner, Window Panels and Side Panels. | | In accordance with Arkansas Aerospace Inc Drawing List 1000-016, dated 06/04/92, or later approved revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 21.02.2007 |
| 10015724 | RAYTHEON AIRCRAFT COMPANY | EASA.IM.A.S.01733 | | FAA STC SA8411SW | BAE 125-1000A | Installation of Electronic Equipment Racks. | | In accordance with Arkansas Aerospace Inc Drawing List 1000-020, dated 06/21/92, or later approved revisions. | | 21.02.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|--|---|---------------------|--|--|---------------|
| 10042983 | RAYTHEON AIRCRAFT COMPANY | EASA.IM.A.S.01768 | | FAA STC ST01331WI-D | HAWKER 800XP | Installation of Collins Integrated Flight Information System (IFIS 5000) and Pro Line 21 Radios | | Collins Master Data List AAC-0360-001 Rev., dated July 8, 2004, or later approved revision. Collins Airplane Flight Manual Supplement ref. AAC-0366-001 Rev., dated June 6, 2005, or later approved revision. Collins Maintenance Manual Supplement ref. AAC-0 (...) | 1. Raytheon Aircraft Company Kit 149-3419-0001 is prerequisite prior installation and approval of STC (EASA.IM.A.S.01768). 2. Raytheon Aircraft Company Kit 149-3420-0001 is prerequisite prior installation and approval of STC (EASA.IM.A.S.01768). 3 (...) | 26.01.2006 |
| 10015448 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01191 | | FAA STC SA01129AT-D | RAYTHEON 1900C | Installation and operation of a Lockheed Martin/Fairchild FDR F1000, solid State flight data Recorder installed. Master Drawing List MDL9606 Initial Revision, dated January 10, 1997 or later FAA approved revision. | | Master Drawing List MDL9606 Initial Revision, dated January 10, 1997 or later FAA approved revision. | 1. FAA approved Airplane Flight Manual Supplement AFMS9606, dated 18 APR 1997 or later approved revision, is a required part of this STC. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in t (...) | 31.08.2005 |
| 10015506 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01384 | | FAA STC SA4899SW | BOMBARDIER CL-600-2B16 | Installation of Wulfsberg Filefone IV/VI Communication System | | -Arkansas Modification Center Inc Drawing List No 600-028 rev D2 dated October 9 1990 or later approved revision-AFM Supplement dated January 26 1983 or January 25 1988 or later approved revision for Filefone IV or AFMS dated September 15 1990 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.07.2005 |
| 10015507 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01385 | | SA5953SW | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 | FAA STC SA5953SW. Installation of a nose gear door ground release mechanism | | Arkansas Modification Center Inc Drawing List No 600-076 rev F dated July 22 1988 or later approved revision | | 25.07.2005 |
| 10015508 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01386 | | SA7457SW | BOMBARDIER CL-600-2B16 | FAA STC SA7457SW. Installation of Sperry LSZ-850 Lightning Sensor System | | -Arkansas Modification Center Inc Drawing List No 600-074 rev A dated September 15 1988 or later approved revision-AFM Supplement dated July 22 1988 is required | | 25.07.2005 |
| 10015509 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01387 | | FAA STC SA7583SW | BOMBARDIER CL-600-2B16 | Installation of Sperry Primus 870 Weather Radar System | | -Arkansas Modification Center Inc Drawing List No 600-073 dated March 20 1989 rev A dated April 12 1989 or later approved revision-AFM Supplement dated April 14 1989 or later approved revision | | 25.07.2005 |
| 10015510 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01388 | | FAA STC SA7871SW | BOMBARDIER CL-600-2B16 | FAA STC SA7871SW. Installation of air driven generator inspection window | | Arkansas Modification Center Inc Drawing List No 600-102 rev A dated August 12 1990 or later approved revision | | 25.07.2005 |
| 10015511 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01389 | | FAA STC SA7878SW | BOMBARDIER CL-600-2B16 | Installation of a DPI Labs 1800 Membrane Switch Control System | | Arkansas Modification Center Inc Drawing List No 600-093 rev A3 dated September 12 1990 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 25.07.2005 |
| 10015512 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01390 | | FAA STC SA7881SW | BOMBARDIER CL-600-2B16 | FAA STC SA7881SW. Installation of Sensor S67-1572 Global Positioning Antenna | | Arkansas Modification Center Inc Drawing List No 600-105 rev A dated August 30 1990 or later approved revision | | 25.07.2005 |
| 10015513 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01391 | | FAA STC SA7880SW | BOMBARDIER CL-600-2B16 | Installation of an alternate location for the fuel panel | | Arkansas Modification Center Inc Drawing List No 600-101 rev A2 dated September 07 1990 or later approved revision | | 25.07.2005 |
| 10015514 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01392 | | FAA STC SA7873SW | BOMBARDIER CL-600-2B16 | FAA STC SA7873SW. Installation of a cockpit Overhead Panel Extension | | Arkansas Modification Center Inc Drawing List No 600-099 rev A1 dated August 30 1990 or later approved revision | | 25.07.2005 |
| 10015515 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01393 | | FAA STC SA7872SW | BOMBARDIER CL-600-2B16 | FAA STC SA7872SW. Installation of a cockpit Center Pedestal Extension | | Arkansas Modification Center Inc Drawing List No 600-100 rev A1 dated August 31 1990 or later approved revision | | 25.07.2005 |
| 10015516 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01394 | | FAA STC SA7869SW | BOMBARDIER CL-600-2B16 | FAA STC SA7869SW. Installation of access door | | Arkansas Modification Center Inc Drawing List No 600-104 rev A dated August 6 1990 or later approved revision | | 25.07.2005 |
| 10015517 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01395 | | FAA STC SA7870SW | BOMBARDIER CL-600-2B16 | FAA STC SA7870SW. Installation of a Rosen Sun Visor System | | Arkansas Modification Center Inc Drawing List No 600-103 rev A dated August 12 1990 or later approved revision | | 25.07.2005 |
| 10015518 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01396 | | FAA STC SA7879SW | BOMBARDIER CL-600-2B16 | Installation of a cabin emergency lighting system | | Arkansas Modification Center Inc Drawing List No 600-097 rev A2 dated September 10 1990 or later approved revision | | 25.07.2005 |
| 10015519 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01397 | | FAA STC SA7886SW | BOMBARDIER CL-600-2B16 | Installation of a cabin oxygen system | | Arkansas Modification Center Inc Drawing List No. 600-098 Revision A-3 dated September 11 1990 or later approved revision. | | 11.08.2005 |
| 10015520 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01398 | | FAA STC SA7887SW | BOMBARDIER CL-600-2B16 | Installation of a Cabin Water System | | Arkansas Modification Center Inc Drawing List No 600-096 rev A2 dated September 05 1990 or later approved revision | | 25.07.2005 |
| 10015521 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01399 | | SA3703SW | BOMBARDIER CL-600-2B16 | FAA STC SA3703 SW. Installation of Thrust Reverser Access Panel in the Lower Nacelle | | Arkansas Modification Center Inc Drawing List No 600-7966 dated April 15 1986 or later approved revision | | 25.07.2005 |
| 10015522 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01400 | | FAA STC SA4900SW | BOMBARDIER CL-600-2B16 | FAA STC SA4900SW. Installation of Dual Baker M1045 Cockpit Audio Systems with a Baker M1050 Cabin Paging/Chime Amplifier | | -Arkansas Modification Center Inc Drawing List No 600-030 rev E dated July 20 1988 or later approved revision-AFM Supplement dated January 26 1983 rev A dated August 3 1988 | | 25.07.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|--|---|---------------------|--|--|---------------|
| 10015523 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01401 | | FAA STC SA7460SW | BOMBARDIER CL-600-2B16 | Installation of Honeywell CG1230AC10 Lasertrack Navigation System | | -Arkansas Modification Center Inc Drawing List No 600-086 rev C dated January 23 1990 or later approved revision-AFM Supplement dated July 22 1988 rev B dated February 2 1990 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 25.07.2005 |
| 10015524 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01402 | | FAA STC SA5718SW | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 | Installation of Airshow 100/200 Cabin Video System | | Arkansas Modification Center Inc Drawing List No 600-045 rev J3 dated September 10 1990 or later approved revision | | 25.07.2005 |
| 10015525 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01403 | | SA5830SW | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 | FAA STC SA5830SW. Installation of Radio Master Switch System | | -Arkansas Modification Center Inc Drawing List No 600-062 rev C dated July 15 1988 or later approved revision-AFM Supplement dated July 22 1988 required for -2B16 | | 25.07.2005 |
| 10015526 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01404 | | SA7451SW | BOMBARDIER CL-600-2B16 | Installation of a Dual EFIS Master Switch System | | -Arkansas Modification Center Inc Drawing List No 600-082 rev A dated July 15 1988 or later approved revision-AFM Supplement dated July 22 1988 is required | | 25.07.2005 |
| 10015527 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01405 | | FAA STC SA7454SW | BOMBARDIER CL-600-2B16 | Installation of Shepherd Security System | | Arkansas Modification Center Inc Drawing List No 600-076 rev B1 dated September 13 1990 or later approved revision | | 25.07.2005 |
| 10015528 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01406 | | FAA STC SA7459SW | BOMBARDIER CL-600-2B16 | Installation of a Dual FMS Master Switch System | | -Arkansas Modification Center Inc Drawing List No 600-083 rev A dated July 16 1988 or later approved revision-AFM Supplement dated July 22 1988 is required | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 25.07.2005 |
| 10015552 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01446 | | FAA STC ST00141AT | BEECH JET 400A | Installation of a Rockwell Collins TCAS 4000 | | -FAA STC ST00141AT-Raytheon Aircraft Services Master Drawing list No 12436-0, Rev E dated Dec. 26, 2005 or later EASA approved revisions-Airplane Flight Manual Supplement/TCAS 4000, doc. No 12436-16 dated 6/5/01 or later EASA approved Revisions-Inst (...) | Certification Basis: FAA TADS A16SW and Leaflet No 8 Rev 2 | 25.01.2006 |
| 10015655 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01631 | | FAA STC ST02599AT | HAWKER 700A AND 700B | Installation of a "Upgraded Height Keeping Equipment that complies with Reduced Vertical Separation Minimum (RVSM) Criteria". | | In accordance with Master Drawing List MDL RAF02101, Revision 2, dated January 8, 2003 or later approved revisions.FAA approved Airplane Flight Supplements, Document Numbers FMS-RAF02101A and if applicable Document Number FMS-RAF02101B, dated Decemb (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 19.12.2005 |
| 10015776 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01821 | | FAA STC SA01366W1 | BEECH 1900C, 1900D | Installation of an ACSS T2CAS in Raytheon Aircraft (Beech) 1900C and 1900D | | Master Drawing List, Report No. GEN-041115-002,Revision A, dated 24-May-2005, or later approved revisionAirplane FlightManual Supplement, Report No GEN-041115-020, Rev. #IR#, dated 21-Jul-2005, or later approved revisionInstructions for Continued Ai (...) | Installation must be done subsequent to or in conjunction with EASA STC EASA.IM.A.S.00762 (FAA STC SA00560DE) | 28.11.2006 |
| 10015777 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01822 | | FAA STC SA01366W1 | RAYTHEON (BEECH) 1900D | Installation of an IS&S Air Data Display Unit on copilot side in Raytheon Aircraft (Beech) 1900D. | | RAS Master Data List RAS1900E020, Rev. D, dated 31-Oct-2005, or later FAA approved revisionAirplane Flight Manual Supplement, RAS1900E022, Rev. -, or later approved revisionInstructions for Continued Airworthiness, RAS1900E024, Rev. B or later released revision. | | 20.10.2006 |
| 10015835 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.01926 | | FAA STC SA00349W1 | BEECH 1900D | Installation of a right hand forward lavatory. | | In accordance with United Beechcraft Inc Drawing Namber RA5009357-1, dated July 28, 1995, or later approved revisions.FAA approved United Beechcraft Inc Aircraft Flight Manual Supplement Manual No. RA5009357FM1, dated July, 1995 or later approved revisions. | None | 28.03.2006 |
| 10016056 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.02233 | | FAA STC SA01404W1 | BEECH 200, 200C, B200C, BEECH 300, B300C | Installation of Collins IFIS 5000 | | FAA approved RAS Master Data List, Doc. No. RAS050600021, Revision 5.0, dated 30-Aug-2006 or later approved revisionFAAapproved Airplane Flight Manual Supplement, Doc. No. RAS050600034, dated 19-Sep-2006, or later approved revisionInstructions for C (...) | This change is limited to airplanes equipped with Collins Pro Line 21 System and includes the FSU-5010 File Server Unit, CCP-3000 Cursor Control panel and either the XMWR 1000 XM Weather Receiver or VHF 4000/CMU4000 Universal Weather System and associated antenna. | 15.01.2007 |
| 10016151 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.02353 | | FAA STC ST03379AT | 400, 400A, 400T | Replacement of a Standby Altimeter installation in the Raytheon (Beech) models 400, 400A, 400T aircraft in accordance with Master Data List (MDL), RAS1106DL004, Revision 1, dated, January 4, 2007, or laterFAA Approved Revision. | | see above | This approval should not be extended to other aircraft of this model on which other previously approvedmodifications are incorporated unless it is determined by the installer thatthe interrelationship between thischange and any of those other previo (...) | 25.04.2007 |
| 10016166 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.02393 | | FAA STC SA01396W1 | BEECH 390 | Modification to install a Collins IFIS 5000, a file server unit, cursor control panel, XM weather receiver and antenna | | RAS Master Data List: RAS050500169, Rev 2.0, dated 20 June 2006 or later FAA approved RevisionFAA approved RAS AFM-S Doc. No. RAS050500183, dated 7 July 2006, or later FAA approved Revision | | 03.07.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|------------------------------|--|---|---|---|---------------|
| 10016167 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.02394 | | FAA STC SA01426W1 | C90A, C90GT | Installation of replacement static inverters. Data required: 1. RAS Master Data List RAS230600970, Revision 1.0, dated January 30, 2007, or later approved revision. | | see above | 1. Compatibility of this design change with previously approved modifications must be determined by the installer. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person wri (...) | 03.05.2007 |
| 10042555 | RAYTHEON AIRCRAFT SERVICES, INC. | EASA.IM.A.S.00352 | | | BEECH 1900C | Installation of a cargo web "Z" net in the forward fuselage and loaders chair. The "Z" net and loaders chair cannot both be installed in the airplane at the same time. Installation of the two configurations are per Raytheon Aircraft Services Top drawi (...) | | | 1. The FAA approved Airplane Flight Manual Supplement p/n 1900C90021, dated November 6, 1996, or later FAA-approved revision is required equipment, when this STC is installed. 2. Compatibility of this design change with previously approved modifications (...) | 07.12.2004 |
| 10015915 | RCM NORMALIZING | EASA.IM.A.S.02045 | | SA5202NM | COMMANDER AIRCRAFT 112 | Validation of FAA STC SA5202NM: Installation of Lycoming engine model IO-360-C1D6 modified in accordance with STC SE5203NM & Propeller Hartzell HC-E3YR-1RF/F7392. | | AFM Supplement No. 1, dated 2 february 1991 and Supplement No. 2, dated 23 February 1998. Installation Manual, Installation Drawings, Parts List, RCM-1000. | See AFM Supplement No. 1. | 25.06.2007 |
| 10016700 | RCM NORMALIZING | EASA.IM.E.S.01014 | | FAA STC SE5203NM | LYCOMING ENGINES IO-360-C1D6 | Installation of turbocharger and induction air intercooler in accordance with #Drawing List Report No. 1# dated August 20, 1990, or later FAA approved revision. | | Installation Instructions Continued Airworthiness Instructions | Compatibility of this design change with previously approved modifications must be determined by the installer. The installation should not be incorporated in any engine unless its determined that the interrelationship between this installation and (...) | 31.05.2007 |
| 10014230 | RECTIMO AVIATION | EASA.A.S.01393 | | | PIPER PA-28-181, ARCHER III | GNS 430 installation with MD 41-1488 (HSI and AP coupled) Installation of GNS 430 and MD 41-1488 coupled with KI 525A | | FACE RECTIMO AVIATION 2005/05-0-FGSMDFAMS "Supplément au manuel de vol approuvé EASA concernant l'utilisation de l'avion équipé d'un GPS GARMIN GNS 430; Edition Juillet 2007" | Limited to S/N 2843066 F-GSMDGPS use is forbidden for BRNAV and approach Auto Pilot coupling is forbidden under 2000 fAGL | 20.07.2007 |
| 10015015 | RECTIMO AVIATION | EASA.A.S.03319 | | | PARTENAVIA P68C | Installation of a COMNAV/GPS GARMIN GNS 430 (BRNAV without coupling to the auto-pilot) a COMNAV GARMIN SL 30, a stormscope L3 WX 500 and an ELT ARTEX ME 406. | | Modification Approval Sheet ref. 2007/40 dated October 8, 2007 Airplane Flight Manual Supplement ref. "Edition Octobre 2007" dated October 8, 2007. | This STC is limited to the S/N 279. | 22.01.2008 |
| 10015112 | RECTIMO AVIATION | EASA.A.S.03459 | | | PIPER PA-28R-201T | Replacement of : King KX155, KI209 and Garmin GPS 155 by Garmin GNS430 coupled to HSI and autopilot. Installation of a RMI KI 229. Removal of King KX155, KI209 and Garmin GPS 155 Installation of King RMI KI 229, Garmin GNS 430 with MD 41 coupled to HSI KI525A and autopilot. | | FACE Rectimo Aviation 2008/03 AFMS EASA approved Flight Manual Supplement for aircraft equipped with GPS Garmin GNS 430 using GPS GNS 430 in VFR mode coupled to HSI and autopilot, ed. July 2008 | Applicable to S/N 7803232 F-GHYG. GPS use is limited to Day VFR flights in sight of ground or water. | 22.09.2008 |
| 10034509 | REGENT AEROSPACE CORPORATION | | | FAA STC ST02274LA | B777-222 | Interior modifications in accordance with Regent Aerospace Corporation Master Data List MDL-4100, Revision E, dated November 25, 2009. FAA STC ST02274LA refers. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Regent Aerospace Corporation MDL-4100, Revision E Regent Aerospace Corporation CMM 25-21-01, Revision 4 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | Limited to B777-222 aircraft serial number 30214 only | 08.04.2011 |
| 10043975 | REIMS AVIATION INDUSTRIES | | | TCCA SA12-74 | BD-700-1A10, BD-700-1A11 | Installation, Provisions for P/N 444-DMC57-1 Securaplane Antenna. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Transport Canada STC No. SA12-74 Issue 1 dated 24 October 2012, ACS-NAI Ltd. Modification Data Summary number H791000 Revision N/C; Mandatory Maintenance Manual Supplement H791070 revision N/C; or later revisions of the above listed documents approved b (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 08.03.2013 |
| 10016335 | REVUE THOMMEN AG | EASA.IM.A.S.02644 | | FAA STC ST10685SC | LEARJET 60 | Installation of Revue Thommen 3A43 Standby Altimeter. | | As noted in MDL3460003, rev A. | | 12.03.2008 |
| 10016337 | REVUE THOMMEN AG | EASA.IM.A.S.02646 | | FAA STC ST10684SC | LEARJET 45 | Installation of Revue Thommen 3A43 Standby Altimeter. | | As noted in MDL3445002, rev A (or later approved revisions). | | 12.03.2008 |
| 10043917 | RHEINLAND AIR SERVICE GmbH | | | | MT-7-235 | Installation of EDM930 Installation of J.P. Instruments (JPI) Primary Engine Data Management System in accordance with JPI Installation Manual for the EDM 900/930, Report 908, dated September 9, 2004. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:- the following paragraph(s) at a later amendment: CS 23 Amendment 3 for following paragraphs: 23.301, 601, 603, 611, 613, 771, 77 (...) | Airplane Flight Manual Supplement No. 930-0001, Revision C, FAA approved 10.12.2004; or later revisions of the above listed documents approved by EASA - Description of Installation RAS-12-03-02-DO-01, dated 27.1.2013 or later approved Revisions - Ma (...) | This approval shall be limited to aircraft MT-7-235 MSN 18067C. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will (...) | 05.03.2013 |
| 10015602 | ROCKWELL COLLINS AEROSPACE | EASA.IM.A.S.01538 | | ST01205NY | BOMBARDIER CL-600-2B16 | Installation of Head Up Guidance System (HGS) Model 2100 (Phase 2) | | STC Master Drawing List No. 9701-1085 Revision N or later approved revision. AFM Supplement 9701-1087 Revision D or later approved revision. | | 15.08.2005 |
| 10016646 | ROCKWELL COLLINS AEROSPACE | EASA.IM.A.S.03089 | REV. 2 | FAA STC ST01100SE | DHC-8-401 & DHC-8-402 | Updates for Flight Management System (FMS) HGS 4100. FAA acceptance letter 1305-09-387, dated 21.09.2009. | JAR/CS-25 as per EASA.IM.A.191 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original (...) | EASA approved AFM Supplement No. 9701-1192, Rev. C or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 12.04.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|-------------------|----------------|---------------------------------|---|---|--|--|--|---------------|
| 10042948 | ROCKWELL COLLINS AEROSPACE | EASA.IM.A.S.01076 | | FAA STC ST01205NY | CL-600-2B19 | Installation of Head-Up Guidance System (HGS) model 2100 | | -FAA STC ST01205NY-STC Master Drawing List No 9701-1085 Revision L dated November 02,2004.configuration 1000-0450-009 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 11.02.2005 |
| 10043977 | ROCKWELL COLLINS AEROSPACE | EASA.IM.A.S.02968 | | FAA STC ST02177LA-T | CL-600-2B16 | HUD/EVS Model 6605 Validation on CL-605 | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -FAA STC No. ST02177LA, dated 03 August 2012.-Rockwell Collins Master Drawing List for Model 6605 Head-Up Display System, Document No. 9701-1267, Revision H, dated 12 July 2012.-Airplane Flight Manual Supplement (AFMS) for the installation of the HUD (...) | The performance of Model 6605 HUD systems has demonstrated acceptability as a supplementary display of primary information. The EVS has demonstrated that the EVS image does not degrade the safety of the flight or interfere with the required pilot tasks (...) | 08.03.2013 |
| 10015727 | ROCKWELL COLLINS AEROSPACE & | EASA.IM.A.S.01743 | REV. 3 | FAA STC ST01390LA | CL-600-2C10, CL-600-2D15 CL-600-2D24, CL-600-2E25 | Installation of Head-Up Guidance System (HGS) Model 4200. Addition of CL-600-2E25 (CRJ 1000) to HGS 4200 STC | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI A-1000-01-HGS The requirements for environmental protection and the associated certifi (...) | -EASA STC no. EASA.IM.A.S.01743 Revision 1 dated 31 May 2007-FAA STC no. ST01390LA dated 28 December 2010- Rockwell Collins Master Drawing List No. 9701-1099 at Revision N or later approved revisions-Rockwell Collins Airplane Flight Manual Supplement (...) | None | 21.04.2011 |
| 10016677 | ROCKWELL COLLINS AEROSPACE & | EASA.IM.A.S.03193 | | FAA STC ST00129SE | BOMBARDIER DHC-8-100 SERIES BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-300 SERIES | Validation of the FAA STC ST00129SE for the HGS 2800 on the Dash 8-100/200/300 Series Aircraft (editorial change). | | German LSA STC (TA002) Austrian AustroControl STC (ACG.21NE.046) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 27.04.2009 |
| 10015761 | ROCKWELL COLLINS Inc. | EASA.IM.A.S.01797 | | FAA STC ST01903LA | BOEING 777-200 | Total Entertainment System. Index List 1304641 rev new dated 2 August 2006. Modification comprises a software upgrade to the VMOD, re-identification of the VMOD and re-installation in the Video Control Centre. | | FAA STC ST01903LA dated 9 August 2006. IFE VMOD upgrade, ref Rockwell Collins System List 1310328-001 rev D. Continued Airworthiness Document, 130944 rev AF. | Limited to B777-200 models operated by British Airways plc. | 19.01.2007 |
| 10016243 | ROCKWELL COLLINS Inc. | EASA.IM.A.S.02517 | | FAA STC ST02055LA | AIRBUS A330-203 | IFE VMOD upgrade per System List 1310929-001/IFE VMOD Upgrade consists of removing the existing VMOD and re-installing an upgraded VMOD in the Video Control Center (VCC). | | FAA STC ST02055LA System List, Air France A330-203 VMOD Upgrade, ref. 1310929 Rev. A, dated 21.02.2007 or later approved revisions. Index List, Total Entertainment System # A330 Aircraft, ref. 1304681, Rev. NEW or later approved revisions | | 20.11.2007 |
| 10016244 | ROCKWELL COLLINS Inc. | EASA.IM.A.S.02518 | | FAA STC ST02056LA | AIRBUS A340-311 AIRBUS A340-313 | IFE VMOD upgrade per System List 1310930-001/IFE VMOD Upgrade consists of removing the existing VMOD and re-installing an upgraded VMOD in the Video Control Center (VCC). | | FAA STC ST02056LA - System List, Air France A340-311/313 # VMOD Upgrade, ref. 1310930-001 Rev. B, dated 21.02.2007 or later approved revisions - Index List, Total Entertainment System # A340 Aircraft, ref. 1304684, Rev. # or later approved revisions | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 04.10.2007 |
| 10016670 | ROCKWELL COLLINS Inc. | EASA.IM.A.S.03156 | | FAA STC NO. SA5098NM | AIRBUS A320-232 | Installation of Sony Trans Com Inc. Video Entertainment System | | # FAA STC SA5098NM# Sony Trans Com Inc. Index List, ref. doc. 062-0130, Revision AA, dated 12th May 2004 (*)# Instruction for Continued Airworthiness, ref. doc. 070-4328-000, Revision BF, dated 16th December 2005 (*) or later revisions a (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 29.04.2009 |
| 10030815 | ROCKWELL COLLINS INC. | | | FAA STC NO. ST01838LA | B747-400 SERIES | Installation of Rockwell Collins enhanced Total Entertainment System in accordance with System List 1309713-002-004, Revision Y, dated September 8, 2005 or later FAA approved revisions. Validation of FAA STC ST01838LA | The Certification Basis for the original product remains applicable to this certificate/ approval. | Instructions for Continued Airworthiness document nr. 1304944 Aircraft Maintenance Manual Supplement 44-00-750r later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 08.07.2010 |
| 10031359 | ROCKWELL COLLINS INC. | | | FAA STC ST01829LA | B747-400 | Installation of Rockwell Collins enhanced Total Entertainment System in accordance with System List 1304631, Revision N/C, dated August 16, 2005 or later FAA approved revisions. Validation of FAA STC ST01829LA | The Certification Basis for the original product remains applicable to this certificate/ approval. | Instructions for a Continued Airworthiness document nr. 1304944 Rev. AB Aircraft Maintenance Manual Supplement 44-00-750r later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | none | 11.08.2010 |
| 10015340 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.00392 | | FAA STC ST01202WI-D, STC EASA.I | SEE ATTACHED EASA APPROVED TYP | Modification of an existing FAA-approved Collins TPR-720, TPR-900 or TPR-901 Mode S transponder installation in Part 25 aircraft. | | - FAA STC No ST01202WI-D.- Collins Drawing Index as listed on AML No ST01202WI-D, revision original, or later FAA-approved revisions. | 1. Prior installation and approval of a Collins TPR-720, TPR-900 or TPR-901 Mode S transponder is a prerequisite to this STC. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph (...) | 29.03.2005 |
| 10015344 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.00444 | | FAA STC ST00587WI-D | BOEING 747-400 | Modification of Collins ACARS DLM-700B Management Unit (MU) System Installation to Collins ACARS DLM-900 MU System Installation. | | - FAA STC ST00587WI-D- Drawing Index 992-4300-001, no revision, dated 9/25/97 and Maintenance Manual Supplement (MMS) ACC-97-131, No revision, dated 10/17/98. | 1. Collins ACARS DLM-700B Management Unit System Installation, EASA STC EASA.IM.A.S.00444, is required to be installed prior to or concurrent with using this STC. 2. This STC is approved only for the product configuration as defined in the approved d (...) | 21.12.2004 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------|----------------|----------------------|--|---|---------------------|--|---|---------------|
| 10015368 | ROCKWELL COLLINS, INC. | EASA.IM.A.S.0090 | | FAA STC SA5859NM | A320 | Installation of Video Entertainment System in accordance with FAA approved Hugues-Avicom Index list 60Z78, Revision "A", dated September 24, 1992, or later FAA approved revisions. | | | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft configurations shall be determined by the installer. | 09.08.2004 |
| 10015380 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.01062 | | FAA STC ST01713LA | BOEING 747-400 | Installation of Rockwell Collins enhanced Total Entertainment System in accordance with FAA STC No ST01713LA issued February 12, 2005 | | -System List 1309056, Configuration -002, Revision N, dated February 12, 2005 or later FAA approved Revision | 1. FAA STC ST015075E must be installed in conjunction with this installation. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft (...) | 01.03.2005 |
| 10015464 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.01214 | | SA1930CE-D | BEECH B300 BEECH B300C | Installation of Collins TCAS-94 TCAS II System. | | Collins Drawing Index 992-1974-001, no revision, dated 16 October 1991 Airplane Flight Manual Supplements and Maintenance Manual Supplements as follows in annex Table Applicability AFM Supplement Maintenance Manual Supplement TCAS-94 (TCAS II) an (...) | 1. Installation of STC SA1929CE-D, mode S transponder (single or dual) is a prerequisite to this STC. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compat (...) | 20.08.2005 |
| 10015530 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.01409 | | ST00673WI-D | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 | Modification of Collins VHF-22A/VHF-22B VHF COMM Receiver Transmitter with CTL-22 Control Pro Line II System. Installation to a Collins VHF-22/VHF-22D VHF COMM Receiver Transmitter with CTL-22C 8.33 khz Pro Line II System Installation | | Collins Installation Instructions as listed on AML No ST00673WI-D Revision- or later approved revision | | 26.07.2005 |
| 10015531 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.01410 | | ST01142WI-D | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 | Modification of a single or dual Collins TDR-94 or TDR-94D Mode S Transponder System installation with CTL-92 control head to a Flight ID capable Collins TDR-94 or TDR-94D Mode S Transponder installation with CTL-92E control head | | Collins Installation Instructions as listed on AML No ST01142WI-D Revision or later approved revision | | 26.07.2005 |
| 10015615 | ROCKWELL COLLINS, INC. | EASA.IM.A.S.01554 | | FAA STC SA3979NM | B757-200 | Installation of Trans Com Video Projection System | | Trans Com Index List 062-0128 Revision "*" dated March 10 1989 or later approved revision | | 20.06.2006 |
| 10015631 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.01584 | | ST01745CH | HAWKER HS 125-700A | Installation of Rockwell Collins Traffic Alert and Collision Avoidance System (TCAS II) in Raytheon HS 125 Series 700A Aircraft | | -Elliott Aviation Master Data List 2004-576DL, Rev 2, dated December 10, 2002 or later approved revisions-Airplane Flight Manual Supplement Elliott Aviation Document No FS2004-576, Rev 0, FAA Approval dated December 15, 2002, or later approved revision (...) | | 06.12.2005 |
| 10015691 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.01679 | REV. 1 | ST02837AT | BOEING 737-600 BOEING 737-700 BOEING 737-800 BOEING 737-900 | Installation of a Rockwell Collins CMU-900 Communications Management Unit, VHF-920 VDL Mode 2 Radio and the APM-900 Airplane Personality Module to provide Controller Pilot Data Link Communications (CPDLC) Baseline 1 services | | Hollingshead International Master Data List 2755-01, Revision B, dated 27 April 2004, or later approved revision. | In accordance with the 2755-AFM Rev A, dated December 12, 2005, or later EASA approved Revisions. | 31.07.2009 |
| 10015735 | ROCKWELL COLLINS, INC. | EASA.IM.A.S.01758 | | FAA STC SA5135NM | B737-800 | Installation of Audio Video Entertainment System with Rockwell Collins System No. 1309667-100 and 1310244-001 according to FAA STC SA5135NM Remark: Only 1 EASA STC to be issued by Programmes Department to cover both applications (working hours inform (...)) | | - Rockwell Collins Index List 062-0138-000, Revision #W dated 19 January 2006 or later EASA approved Revision- Maintenance Instructions for Continued Airworthiness, Rockwell Collins document 070-4328-000 | The limitations according to FAA STC SA5135NM apply. | 17.08.2006 |
| 10015906 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02032 | | ST00525WI-D | | Modification of Collins 618M-3/618m-4 VHF COMM Receiver Transmitter Installation to a 618M-5 VHF COMM Receiver Transmitter Installation with either a Gables G7406-02 or Avtech 7522-1-2-VHF COMM Receiver Transmitter. | | Collins Installation Instructions as listed on AML No. ST00525WI-D Revision -, or later approved revision. | | 31.07.2006 |
| 10015943 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02085 | | SA2656CE | BOEING 747-100 BOEING 747-100B SUD BOEING 747-200B BOEING 747-200C BOEING 747-200F BOEING 747-300 BOEING 747-SP BOEING 747-SR | Collins traffic alert and collision avoidance system (TCAS II) | | 1. Drawing Index 992-1944-0012. Aircraft Flight Manual Supplement ACC-90-113 | 1. Installation of Collins Mode S Transponder , per STC SA2636CE (P-EASA.IM.A.S.02087) is a prerequisite to this modification. 2. ACC-90-113, Airplane Flight Manual Supplement 3. 992-1944-001, Drawing Index | 01.02.2007 |
| 10015944 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02086 | | SA2676CE | BOEING 747-100 BOEING 747-100B SUD BOEING 747-200B BOEING 747-200C BOEING 747-200F BOEING 747-300 BOEING 747-SP BOEING 747-SR | Collins traffic alert and collision avoidance system (TCAS II) | | Drawing Indexes 992-1978-001, 992-1989-001, and 992-2696-001. Aircraft Flight Manual Supplement ACC-99-042/-43/-44/-45. | Installation of Collins Mode S Transponder, per STC SA2636CE (P-EASA.IM.A.S.02087) is a prerequisite to this modification. Aircraft Flight Manual Supplement ACC-99-042/-43/-44/-45. Drawing Indexes 992-1978-001, 992-1989-001, and 992-2696-001. | 01.02.2007 |
| 10015945 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02087 | | SA2636CE | BOEING 747-100 BOEING 747-100B SUD BOEING 747-200B BOEING 747-200C BOEING 747-200F BOEING 747-300 BOEING 747-SP BOEING 747-SR | Dual Collins TPR-720 Mode S Transponder System. | | Drawing Indexes 992-2623-001, 992-2949-001, 992-1954, 992-2976, 992-1977 or 992-1988. | | 01.02.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------|----------------|----------------------|---|--|---------------------|--|---|---------------|
| 10016013 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02179 | | FAA STC SA2685CE | BOEING DC-9-81 (MD-81) BOEING DC-9-82 (MD-82) BOEING DC-9-83 (MD-83) BOEING DC-9-87 (MD-87) BOEING MD-88 | Installation of Collins Traffic Alert and Collision Avoidance System (TCAS II) | | STC No. SA2685CEDrawing Index DWG No. 992-1952-001 (Rev. J)AFM Supplement ACC-99-066 (dated 30 September 1999) | Installation of Mode S Transponder System STC EASA.IM.A.S.02180 (SA2684CE) or EASA.IM.A.S.02181 (SA3082SO) is a prerequisite to this modification. | 04.01.2007 |
| 10016014 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02180 | | FAA STC SA3082SO | BOEING DC-9-81 (MD-81) BOEING DC-9-82 (MD-82) BOEING DC-9-83 (MD-83) BOEING MD-88 | Installation of Dual Collins TPR-720 Mode S Transponders | | STC SA3082SOMaster Drawing List MDL-91-47 Rev. C. | None.Registration (if applicable):- Serialnumber (if applicable):- | 04.01.2007 |
| 10016015 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02181 | | FAA STC SA2684CE | BOEING DC-9-81 (MD-81) BOEING DC-9-82 (MD-82) BOEING DC-9-83 (MD-83) BOEING DC-9-87 (MD-87) BOEING MD-88 DC-9-81 (MD-81) | Installation of Dual Collins Mode S Transponders. | | STC SA2684CEDrawing Index DWG No. 992-1953-001 Rev. C. | None.Registration (if applicable):- Serialnumber (if applicable):- | 04.01.2007 |
| 10016059 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02236 | | ST01580AT-D | BOEING 747-400 | Installation of radio panels and Collins VHF transceivers with 8.33 khz spacing in accordance with Collins Master Data List MDL-01237, revision M, dated August 9, 2005, or later FAA approved revisions. | | Data package containing the following documents, and additional supporting data:MDL 01237, rev. M: MDL Collins VHF700V/-900B in B747-400.CFG-01296: installation of Collins VHF-700B communications system on B747-400 for Air France.ENG-01654: installa (...) | ¿ Data package containing the following documents, and additional supporting data:¿ MDL 01237, rev. M: MDL Collins VHF700V/-900B in B747-400.¿ CFG-01296: installation of Collins VHF-700B communications system on B747-400 for Air France.¿ (...) | 08.08.2007 |
| 10016060 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02237 | REV. 1 | FAA STC ST02410CH-D | BAE 125 SERIES 800A | Installation of Rockwell Collins Pro Line 21 IDS (Integrated Display System) with IFIS-5000 (Integrated Flight Information System) and Engine Instrument Display. | | Installation in accordance with Duncan Aviation Master Drawing List 060912014, Revision G, dated 4/20/07, or later approved revisions.Aircraft to be operated in accordance with Aircraft Flight Manual Supplement, 060912018, Revision Ø dated11/8/07. o (...) | | 28.02.2008 |
| 10016077 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02261 | REV. 1 | FAA STC ST01901LA | BOEING 767-200 BOEING 767-300 | Installation of ARINC 718A MODE S Transponder Activation ofElementary and Enhanced Surveillance Capabilities. | | Hollingshead International Master Data List 2869-01, Revision B, 13/10/2006.AFM Supplement, Hollingshead International Document 2869-AFM, Revision NC, 17/10/2006. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 30.06.2008 |
| 10016202 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02444 | | FAA STC SA01267WI-D | BEECH 300 BEECH B300 BEECH B300C | Installation of Collins Pro Line 21 Avionics System. | | Installation in accordance with:-Collins Master Drawing List ref RCA-0020-001, Revision G, dated 23 July 2008, or laterapproved revisions.Operation in accordance with:- Collins Flight Manual Supplement ref RCA-0030, Revision C, dated 21 July 2008, or (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 02.10.2008 |
| 10016212 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02460 | | FAA STC ST01119WI-D | CESSNA 750 FALCON 10 FALCON 900 FALCON 900EX GULFSTREAM IV GULFSTREAM 1159 (G-II) GULFSTREAM 1159 A (G-II) GULFSTREAM 1159 B (G-II) HAWKER HS 125-700A HAWKER HS 125-700B MYSTERE FALCON 50 | Modification of Collins TDR-94 or TDR-94D Mode S Transponder System installation with stand-alone Collins TRU-4210 or RTU-4220 Radio Tuning Unit(s) to a Flight ID capable Collins TDR-94 or TDR-94D Mode S Transponder System installation with stand-alone Collins RTU-4210 or RTU-4220. | | In accordance with Collins Installation Instructions as listed on AML N°. ST01119WI-D, Revision-, or later approved revision. | (1) Approval of this change in type design applies to the above model aircraft only.(2) Compatibility of this design change with previously approved modifications must be determined by the installer.(3) A copy of this Certificate and FAA Approved Mo (...) | 26.10.2007 |
| 10016249 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02526 | | FAA STC ST00040WI-D | AIRBUS A320-111 AIRBUS A320-211 AIRBUS A320-212 AIRBUS A320-231 AIRBUS A320-232 | Installation of Collins Traffic Alert and Collision Avoidance System (TCAS II). | | FAA STC ST00040WI-DCollins Drawing Index, ref. Doc 992-1979-001, Rev. N, Dated 01.10.2007*Collins Airplane Flight Manual Supplement (AFMS), ref. Doc. ACC-02-009, Rev. A, dated 01.12.03*Collins Maintenance Manual Supplement (MMS) ref. Doc ACC-98-013, (...) | Installation of a Collins TPR-720, TPR 900 or TPR-901 Mode S Transponder is a prerequisite to this modification. | 15.10.2007 |
| 10016273 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02558 | | FAA ST00392AT | BOEING 767-200 SERIES BOEING 767-300 SERIES | Installation of Collins Dual TPR-720 Mode S Transponders. | | Collins Master Drawing List, MDL-93-38, rev. F, 22/03/2000. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 02.11.2007 |
| 10016275 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02562 | REV. 1 | FAA STC ST01470LA-D | 737-600, 737-700 737-800, 737-900 | Installation of Rockwell Collins WXR-2100 Multi-scan (TM) Weather RadarSystem(Revision to the existing STC EASA.IM.A.S.02562 to change the STC holdername and to add more models and correct the equipment reference name) | l.a.w. TCDS | Hollingshead International Inc. Master Data List No 2736-01,Revision A,dated 28-03-04, or later approved revision. | None. | 08.07.2008 |
| 10016287 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02582 | REV. 1 | FAA STC ST01927LA | BOEING 757-200 BOEING 757-300 | Installation of ARINC Characteristics 718A for Mode S Transponder Activation of Elementary and Enhanced Surveillance Capabilities. | | # Hollingshead International Master Data List 2894-01, Revision NC, 10th October 2006.# AFM Supplement 2894-AFM, Revision Original, 24th October 2006. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 15.10.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------|----------------|----------------------|---|--|---------------------|---|---|---------------|
| 10016378 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02702 | | FAA STC ST01454WI-D | BOMBARDIER CL-600-2B19 BOMBARDIER CL-600-2C10 BOMBARDIER CL-600-2D15 BOMBARDIER CL-600-2D24 | Modification of a Collins Control Display Unit, Installation, Type CDU-4100 | | FAA Statement of Compliance ref. L115W-08-522 dated 21 July 2008 FAA STC No. ST01454WI-D dated 25 September 2007/RockwellCollins Drawing Index No. RCA-0113-001, Rev., dated July 31, 2007, or later approved revision. | 1. Prior installation and approval of a Collins Control Display Unit, type CDU-4100, is a prerequisite to this STC. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any (...) | 28.07.2008 |
| 10016379 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02703 | | FAA STC ST01431WI-D | SEE ATTACHED EASA APPROVED TYP | Modification of a Collins DBU-4000 or DBU-4100 Data Base Unit installation to a Collins DBU-5000 installation | | See associated EASA Approved Model List (AML) No. EASA.IM.A.S.02703, initial revision or later EASA approved revisions. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 15.10.2008 |
| 10016395 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02724 | | FAA STC ST01452WI-D | | Modification of Collins FMS-5000 or FMS 6000 and installation in Part 25 Airplanes i.a.w. Drawing index as listed in AML Nr. ST01452WI-D Rev. pr mater approved revisions approved by EASA i.a.w. EASA ED decision 2004/04/CF (or subsequent revision of this decision). | | Approved model List No. ST01452WI-D Installation instructions RCA-2447. | A prior installation and approval of a Collins FMS type FMS-5000 or FMS 6000 is a prerequisite to this STC. Refer to Collins Installation Instructions RCA-2447 in its latest revision for approved FMC substitutes. | 25.03.2008 |
| 10016396 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02725 | | FAA STC ST00643WI-D | BOEING 747-200B BOEING 747-200C BOEING 747-200F BOEING 747-300 | Modification of Collins 618M-3A VHF COMM R/T System Installation to Collins VHF-900B VHF COMM R/T System. | | Data package containing the following documents, and additional supporting data: 1. Collins Drawing Index 992-4304-001 2. FAA STC ST00643WI-D 3. Rockwell-Collins document R98047: STC Certification Plan for Collins VHF-900B 8.33 kHz VHF Communication (...) | This STC is approved only for the product configuration as defined in the approved design data. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 20.03.2008 |
| 10016397 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02726 | | FAA STC SA01430WI-D | BEECH 390 BEECH B200 BEECH B200C BEECH B300 BEECH B300C PIAGGIO P 180 SEE ATTACHED EASA APPROVED TYP | Modification of a Collins DBU-4000 or DBU-4100 Data Base Unit installation to a Collins DBU-5000 Installation. | | Installation in accordance with:-Collins Master Drawing List ref AML No EASA.IM.A.S.02726, dated 15 October 2008, or later approved revision | 1. Prior installation of Collins DBU-4000 part numbers 822-9865-001/-002 or DBU-4100 part numbers 822-0014-101/-102/-103/-104 is a prerequisite to this STC. 2. Prior to installation of this modification the installer must determine that theirrelat (...) | 15.10.2008 |
| 10016398 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02727 | | FAA STC SA01453WI-D | BEECH 300 BEECH 390 BEECH B200 BEECH B300 BEECH B300C CESSNA 525 CESSNA 525A CESSNA 525B PIAGGIO P 180 | Modification of a Collins Flight Management System type FMS-3000 Installed in Part 23 Aircraft. | | Installation in accordance with Collins Drawing Index as listed in AML No EASA.IM.A.S.02727 at latest approved revision. Operation in accordance with Collins Flight Manual Supplement as listed in AML No EASA.IM.A.S.02727 at latest approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 06.11.2008 |
| 10016517 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02888 | | FAA STC SA01434WI-D | BEECH 390 BEECH B200 BEECH B200C CESSNA 525 CESSNA 525A CESSNA 525B PIAGGIO P 180 | Modification of Collins GPS-4000 or GPS-4000A Global Positioning Sensor (GPS) system installation to a Collins GPS-4000S Global Positioning Sensor (GPS) system installation | | Collins Drawing Index RCA-0124-001 Rev-A, dated February 14th, 2007, or later FAA approved revisions AFM Supplement ACC-07-009 Rev-A dated April 2nd, 2007 or later FAA approved revisions | This STC does not approve WAAS LPV Prior FAA approved installation of Collins GPS-4000 part numbers 822-0931-001/-002/-003 or GPS-4000A part number 822-1377-001 is a prerequisite to this STC Prior to installation of this modification the installer mus (...) | 04.02.2009 |
| 10016518 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02889 | | FAA STC ST01433WI-D | BEECH 400 BOMBARDIER BD-100-1A10 (CHALLE) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 (CL-604) BOMBARDIER CL-600-2C10 BOMBARDIER CL-600-2D24 (SERIES) CESSNA 560 FALCON 2000 FAN JET FAN JET FALCON SERIES C FAN JET FALCON SERIES D FAN JET FALCON SERIES E FAN JET FALCON SERIES F FAN JET FALCON SERIES G GULFSTREAM 200 GULFSTREAM G150 HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP LEARJET 60 MYSTERE FALCON 20-C5 MYSTERE FALCON 20-D5 MYSTERE FALCON 20-E5 MYSTERE FALCON 20-F5 MYSTERE FALCON 50 | Modification of Collins GPS-4000 or GPS-4000A Global Positioning Sensor installation to a Collins GPS-4000S installation | | FAA STC ST01433WI-D, Collins Drawing Index, ref. Doc. RCA-0125-001, Rev. A, dated January 18th 2008 or later revisions approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision), Airplane FI (...) | | 24.07.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------|----------------|-------------------------|--------------------------------|--|--|---|--|---------------|
| 10016654 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.03103 | | | SEE ATTACHED EASA APPROVED TYP | Collins CMU-900 ACARS Core Software Upgrade. This modification removes the existing CMU-900 and installs the same part number CMU-900 with updated core software. | | Drawing index ref : RCA-0048-001 revision C, dated 22 September 2008 (or any EASA approved revision) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The i (...) | 27.02.2009 |
| 10035296 | ROCKWELL COLLINS, Inc. | | | FAA STC ST01886LA | B767-300 | Installation of Rockwell Collins CMU-900 Communications Management System, APM-900 Airplane Personality Module, IDC-900 Integrated Datalink Controller & Teledyne TP-4499 Printer. | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this approval. | Hollingshead International Master Data List 2853-01, Revision NC, 04/04/2006 Flight Manual Supplement, Document No. 2853-AFM, Revision NC, 12/05/2006 for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 20 (...) | none | 14.06.2011 |
| 10037095 | ROCKWELL COLLINS, Inc. | | REV. 1 | FAA STC ST01957AT-D | B747-400 | Installation of Collins SAT-906 Satcom Systems provisions in accordance with Timco Master Data List MDL 99T045L001, Rev. C, dated October 21, 1999 or later approved FAA revisions. | The Certification Basis for the original product remains applicable to this certificate/ approval. | Rockwell Collins RCA-280 Rev. - dated March 5, 2010 Instructions for continued airworthiness, Damage Tolerance Inspection. Timco Maintenance Manual Supplement to Chapter 25-00-00-001, September 15, 1999 or later revisions of the above listed documents (...) | None. | 13.04.2012 |
| 10037522 | ROCKWELL COLLINS, Inc. | | | FAA STC NO. ST02177LA | CL-600-2B16 | HUD/EVS Model 6605 Validation of HUD/EVS in a Deactivated State | EASA.IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. ST02177LA dated 04 December 2009. Installation of Rockwell Collins Model 6605 Head-Up Display (HUD) and Enhanced Vision System (EVS) CMA-2600i Infrared Sensor System (ISS) in accordance with FAA Approved Rockwell Collins Master Drawing List (...) | Restricted to ASN 5872. This approval is for the installation and deactivation of the HUD and EVS in accordance with section 9 of the Rockwell Collins Master Drawing List 9701-1267 Revision D-1. | 02.12.2011 |
| 10041711 | ROCKWELL COLLINS, Inc. | | REV. 1 | FAA STC NO. ST01514WI-D | CL600-2B19 REGIONAL JET A21EA | Installation of Rockwell Collins FMS-4200 to provide WAAS/LP Capability. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. ST01514WI-D dated 10 April 2009. Master Drawing List RCA-0179-001, Revision -, dated 8 October 2008. Airplane Flight Manual Supplement ACC-08-025, Revision -, dated 17 December 2008 or later revisions of the above listed documents approved by (...) | Installation and approval of WAAS compatible GPS antennas is required prior to or concurrent with this modification. This STC approves the installation of DBU-5000 Part Number 822-2215-602 (Flip Door/Gray/No AFIS/Ethernet). Installation and approval of a (...) | 12.10.2012 |
| 10041712 | ROCKWELL COLLINS, Inc. | | | FAA STC NO. ST01505WI-D | CL-600-2B16 (CL-604 VARIANT) | Installation of Rockwell Collins FMS-6000 to provide SBAS LPV Capability. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. ST01505WI-D dated 05 November 2010. Master Drawing List RCA-0150-001, Revision H, dated 1 October 2010. Airplane Flight Manual Supplement ACC-09-107, Revision B, dated 19 May 2010 or RCA-0289, Revision -, dated 01 October 2010. Instructions (...) | Restricted to Serial Numbers 5301-5699, 5701 and subsequent. Installation and approval of WAAS compatible GPS receivers/antennas are required prior to or concurrent with this modification. DBU-4000 and DBU-4100 Data Base Units are incompatible with the (...) | 09.10.2012 |
| 10042939 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.01061 | REV. 1 | FAA STC ST01256WI-D | SEE EASA APPROVED MODEL LIST | Modification of a Collins TPR-720, TPR-900 or TPR-901 Mode S transponder system installation with an ATC/TCAS control panel to an Elementary Surveillance compliant Collins TPR-901 Mode S transponder system installation with a flight ID/ATC/TCAS control (...) | | FAA STC ST01256WI-D, Approved model list No ST01256WI-D, Collins Drawing Index RCA-0010-001, Revision E or later approved revisions. | 1. In lieu of an AFM Supplement, the required Gables Flight ID/ATC/TCAS control panel pilot's guide, as applicable per FAA approved Collins Drawing Number RCA-2018-001. Installation Instructions, Mode S Transponder and Flight ID/ATC/TCAS control panel (...) | 06.06.2006 |
| 10042950 | ROCKWELL COLLINS, INC. | EASA.IM.A.S.01114 | | | B737-400, B737-700, B737-800 | Installation of a Video Entertainment System | | Sony Trans Com Inc. Index List 062-0138, Rev New, or later FAA approved revisions | 1. B737-400: For Certification basis refer to FAA TCDS A16W Eplus JAA TGL No 17, IFE Systems 2. B737-700, -800: For Certification basis refer to EASA TCDS JAA/25/97-018 plus JAA TGL No 17, IFE Systems 3. This STC is approved only for the product configuration 1. | 07.04.2005 |
| 10042974 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.01213 | | | BEECHCRAFT B300, B300C | Installation of dual Collins TDR-94/94D Mode S Transponder or installation of single Collins TDR-94/94D Mode S Transponder and single TDR-90 Mode C Transponder on Beechcraft B300 & B300C | | Dual TDR-94/94D: Collins Drawing Index 992-1972-001 and later approved revisions Single TDR-94/94D and Single TDR-90: Collins Drawing Index 992-1973-001 and later approved revisions | 1. Any bottom doubler-antenna is required pre modification. 2. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. 3. This STC is approved (...) | 30.05.2005 |
| 10042991 | ROCKWELL COLLINS, Inc. | EASA.IM.A.S.02261 | REV. 1 | FAA STC ST01901LA | B767-200/-300 | Installation of ARINC 718A MODE S Transponder Activation of Elementary and Enhanced Surveillance Capabilities. | | Hollingshead International Master Data List 2869-01, Revision B, 13/10/2006 AFM Supplement, Hollingshead International Document 2869-AFM, Revision NC, 17/10/2006. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The ins (...) | 31.07.2009 |
| 10046017 | ROCKWELL COLLINS, Inc. | | | FAA STC ST01547WI-D | 750/800XP/850XP/900XP | Modification of Rockwell Collins FMS-6000 System Installation to provide SBAS/LPV Capability | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Rockwell Collins Master Drawing List (MDL) WAAS (SBAS) LPV in Pro Line 21 equipped Hawker Aircraft ref RCA-0166-001 Revision A, dated August 10, 2009. Operation in accordance with Rockwell Collins Airplane Flight Manual Sup (...) | DBU-4000 and DBU-4100 Data Base Units are incompatible with the FMC-6000 (Flight Management Computer) installed by this STC. Loading of the FMS Data base requires PCD-3000, CPAS, or a DBU-5000. If DBU data loading capability is required or desired, Rockw (...) | 06.08.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|-------------------|----------------|--------------------------------|--|---|--|--|---|---------------|
| 10046045 | ROCKWELL COLLINS, Inc. | | | FAA STC SA01636WE-D | SEE EASA APPROVED MODEL LIST | Modification of a Rockwell Collins, Inc. VHF-4000 and VHF-4000E installations | As defined on FAA STC SA01636WI-D The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Rockwell Collins, Inc. Master Drawing List RCA-0359-001 as listed in the AML to this STC or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent r (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 08.08.2013 |
| 10015977 | ROCKY MOUNTAIN AIRCRAFT | EASA.IM.A.S.02129 | REV. 1 | TCCA STC SA02-128 | DHC-6-100, 6-200, 6-300 | Installation of Single or Dual Sandel ST3400 Terrain Awareness Warning System (TAWS)/RMI System | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Revision to EASA STC IM.A.S.02129 validating Transport Canada STCSA02-128 Issue 2. Modification Instruction 181710 Revision M Engineering Data Summary E100300 at initial revision dated 12 March 2010. Drawing 1873419 incorporating DCN RMA1873419/1. Drawin (...) | The audio for dual TAWS configuration must be from a single source in accordance with drawings 1873419, 1873420 and 1873421 incorporating DCNs RMA1873419/1, RMA1873420/2 and RMA1873422/1 respectively. | 17.03.2010 |
| 10016029 | ROCKY MOUNTAIN AIRCRAFT | EASA.IM.A.S.02198 | | TCCA STC SA01-30 | | Removal of Auxiliary Battery | | Compliance Checklist C62766. | 1. An approved Main Battery Installation with a minimum capacity of 34 Ampere-Hours must be embodied as a pre-requisite. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragr (...) | 22.12.2006 |
| 10016030 | ROCKY MOUNTAIN AIRCRAFT | EASA.IM.A.S.02199 | | TCCA STC SA01-17 | BOMBARDIER DHC-6 SERIES 300 | Cabin Evacuation Lighting Control System Change. | | Transport Canada STC SA01-17 Issue 2 Master Data List MDL-1298-01 Aero Design Ltd Flight Manual Supplement FMS396.02 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 10.10.2008 |
| 10016033 | ROCKY MOUNTAIN AIRCRAFT | EASA.IM.A.S.02202 | | FAA STC SA02297NY | BOMBARDIER DHC-6 SERIES 200 BOMBARDIER DHC-6-100 DE HAVILLAND CANADA DHC-6-300 | (1) Installation of a Universal Avionics CVR-30B Cockpit Voice Recorder. (2) Installation of a Fairchild A100S Honeywell SSCVR, or L3 FA2100 Cockpit Voice Recorder. | | (1) Installation in accordance with Amtech Aeronautical Limited Modification, Data Summary MDS977801, Revision 2, dated January 15, 2002 or later approved revisions. (2) Installation in accordance with Field Aviation Company Inc Modification, Instructi (...) | The installation applies only to aircraft not equipped with separate P.A. microphone. Maintenance must be in accordance with Field Aviation Company Maintenance Manual Supplement Report No 199031, Original Issue, dated December 11, 2006, or later approved revisions. | 22.10.2007 |
| 10016286 | ROCKY MOUNTAIN AIRCRAFT | EASA.IM.A.S.02581 | REV. 1 | TCCA STC SA02-119; EASA STC.IM | VIKING AIR LIMITED (DCH) DHC-6 | Dual or Single Garmin GNS530 or GNS430 installation. | | Installation in accordance with Field Aviation Company inc. Modification Instruction 181811, original Issue or later approved revision for single GNS-430 or GNS-530 Installation; 181813, original Issue or later approved revision for single GNS-430W or (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The ins (...) | 21.11.2008 |
| 10016400 | ROCKY MOUNTAIN AIRCRAFT | EASA.IM.A.S.02729 | REV. 1 | TCCA STC SA07-131 | DHC-6-100, DHC-6-200, DHC-6-300 | Installation of single or dual Sandel SN4500 primary navigation display(s) (EHSI) with optional SA4550 primary attitude display(s) (EADI) | FAR 23 | -Installation in accordance with Rocky Mountain Aircraft (RMA) Modification Instruction E100010, Rev., Transport Canada Civil Aviation (TCCA) approved March 10, 2010, or later approved revisions. -Operation in accordance with Rocky Mountain Aircraft R (...) | None | 03.08.2010 |
| 10016511 | ROCKY MOUNTAIN AIRCRAFT | EASA.IM.A.S.02878 | REV. 1 | TCCA STC SA08-38 | BOMBARDIER DHC-6 SERIES 200 BOMBARDIER DHC-6-100 BOMBARDIER DHC-6-300 | Installation of Pitot Heat Indication system. | | Installation in accordance with Transport Canada STC SA08-38, Issue 1, April 07, 2008, or later Transport Canada approved revision. The applicable manuals are as follows: Airplane Flight Manual Supplement : TCCA approved FMS-1270-01, Issue 1, dated 29 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 11.11.2008 |
| 10037908 | ROCKY MOUNTAIN AIRCRAFT | | | TCCA STC SA11-72 | DHC-6 SERIES 100, 200, 300 | Installation of chip detectors on both left hand and right hand engine with caution indication in DHC-6 series 100, 200, and 300 Twin Otter aircraft based on Engineering Data Summary E104600 in accordance with Transport Canada STC SA11-72, Issue 1 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement, Report E104630, Revision 0 Modification Instruction, Report E104610, Revision 0 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this d (...) | None | 05.01.2012 |
| 10038966 | ROCKY MOUNTAIN AIRCRAFT | | | | DHC-6-100/-200/-300 | Installation of Dual or Single Garmin GTN 650 or GTN 750 in DHC-6 series 100, 200, and 300 Twin Otter aircraft based on Engineering Data Summary E102910 & E102911 in accordance with Transport Canada STC SA11-79, Issue 1 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval FAR 23, Amdt. 23-49: 23.1309, 23.1311, 23.1322, 23.1365, 23.1529 FAR 23, Amdt. 23-57: 2 (...) | RMA Flight Manual Supplement E102930, Revision 0 Modification Instruction, Report E102910, Revision 0 (single installation) Modification Instruction, Report E102911, Revision 0 (dual installation) or later revisions of the above listed documents approved (...) | None | 27.03.2012 |
| 10040247 | ROCKY MOUNTAIN AIRCRAFT | | | TCCA STC NO. SA11-80 | DHC-6 SERIES 100 DHC-6 SERIES 200 DHC-6 SERIES 300 | Installation of either Universal Avionics CVR-120A or CVR-120R in accordance with Transport Canada STC SA11-80, Issue 1 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Modification Instruction, Report E103110, Revision 0 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision) Maintenance Manual Supplement, Report E103131, Revision - | none | 19.06.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|-------------------|----------------|----------------------|--|---|--|---|--|---------------|
| 10042836 | ROCKY MOUNTAIN AIRCRAFT | | | TCCA STC SA05-54 | DHC-6 SERIES 100 DHC-6 SERIES 200 DHC-6 SERIES 300 | Installation of a L-3 Skywatch HP TCAS 1 System in accordance with TCCASTC SA05-54 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement, Doc. No. E101430, TCCA approved March 19, 2012. Modification Instruction for Installation of L-3 Communications Skywatch HP Model SKV899, Doc. No. E101410, Rev. A, TCCA approved Sept. 25, 2012 or later revisions of the above (...) | none | 20.12.2012 |
| 10045940 | ROCKY MOUNTAIN AIRCRAFT | | | TCCA SA96-110 | DHC-6 SERIES 300 | Installation of PT6A-34 engines derated to 620 SHP on DHC-6 Series 300 in accordance with TCCA STC SA96-110. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels. | Flight Manual Supplement FMS966401, Issue 1, TCCA approved August 1, 1996. Maintenance Manual Supplement MMS966401, Issue 1, TCCA approved August 1, 1996 or later revisions of the above listed documents approved by EASA in accordance with the Technical Im (...) | Limited to serial numbers 130, 210, 231 and subsequent. The installer must determine whether this design change is compatible with previously approved modifications. If the holder agrees to permit another person to use this certificate to alter the prod (...) | 30.07.2013 |
| 10015218 | ROEDER PRAEZISION GMBH | EASA.A.S.03633 | | | CIRRUS SR20 CIRRUS SR22 | Installation of Avidyne Iridium MLX770 Two Way Datalink. | | - According to Röder Präzision MRL Document No. MR 3400-46-01A, Rev. 3 or later Revisions- To be operated in accordance with approved Röder Präzision AFM-S Document No. 2882AF 3400-46-01A, Original Issue, or later approved Revisions, or To (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 03.12.2008 |
| 10029201 | ROEDER PRAEZISION GMBH | | | | CESSNA P210N | Ferry Tank Provisioning | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Engineering Order 725EO280010-01A-POH Supplement for Cessna P210N, S/N P21000629, Doc. No. 7725 AM 280010 01A, Revision 1, dated 18 February 2010 or later revisions of the above listed documents approved by EASA | Limited to Aircraft S/N P210 00629 | 11.03.2010 |
| 10030847 | ROEDER PRAEZISION GMBH | | REV. 1 | | 206H, T206H | Provisioning for implementation and operation with a "Turtle Pac160-ferry tank" | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-23, Initial Issue (not required by 21A.101, elected to comply by the applicant) The re (...) | - Engineering Order 1704EO280010-01A, dated 26.05.2010-POH Supplement: "Anhang zum Flughandbuch für Cessna 206H and T206H", Doc. No. 1704 AM 2800 10 01A, dated 18.05.2010 or later revisions of the above listed documents approved by EASA | Installation of the ferry tank is not subject of this STC. | 14.07.2010 |
| 10034119 | ROEDER PRAEZISION GMBH | | | | PIPER PA-31-350 | Provision for Camera Installation. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Engineering Order 71335 EO 5300 10 01 Rev. A- Supplement to Pilot Operating Handbook POH 71335 AM 5300 10 01 Rev. A- Instructions for Continued Airworthiness 71335 CA 5300 10 01 Rev. A or later revisions of the above listed documents approved by EASA. | None. | 08.03.2011 |
| 10039546 | ROEDER PRAEZISION GMBH | | | | TU206F | Installation of an additional Alternator with Battery for Mission Purposes | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement 01271AM240030-01A or later revisions of the above listed documents approved by EASA. Instructions for Continued Airworthiness 01271CA240030-01A. Engineering Order 01271EO240030-01A | None | 10.05.2012 |
| 10039554 | ROEDER PRAEZISION GMBH | | | | 172S | Installation of provision for the installation of a camera in the lower fuselage as described in description document 14576DE 5300 10 00D | The Certification Basis of the original product remains applicable to this change. This certificate/ approval involves a change to the certified noise and/or emissions levels. | Master record list Rev A C172S 14576 MR 5300 10 01C (rev.C) dated 10.05.2012. EASA approved Supplement to Pilot's Operating Handbook 14576 AM 5300 1001C (rev.C) dated 27.04.2012 and any later EASA approved revision. Instruction for Continued Airworthiness 14576 CA 5300 10 01B (rev.B) dated 10.04.2012. | Installer must check availability of EASA-approval for one or more referenced STC when airplane is modified per that / those STC(s). This STC is limited to S/N 17258186. This STC provides instructions only for the installation of the provision for a came (...) | 10.05.2012 |
| 10015845 | ROSE UPHOLSTERY, Inc. | EASA.IM.A.S.01945 | | SA10188SC | BAE JETSTREAM 3201 | Cabin Reconfiguration - 12 Passenger Executive Interior | | Rose Aircraft Interiors Inc. Master Drawing Lis, Report No.8265.00.09, Revision 2, dated 04-03-2003 or later approved revision. Instructions for Continued Airworthiness, Revision IR, dated 09-09-2003 or later approved revision. Aircraft Flight Manual Supplement, Report No. 8365.00.10 dated September 2. | | 11.04.2006 |
| 10015376 | ROSEMOUNT AEROSPACE, Inc. | EASA.IM.A.S.01049 | | FAA STC ST02175CH | BOEING 747-400 | Installation of Rosemount Aerospace 8400M Series Cockpit Door Surveillance System | | -CAA-NL report I/VW/LTB/TCP-05.533157 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 06.10.2005 |
| 10015857 | ROSEMOUNT AEROSPACE, Inc. | EASA.IM.A.S.01961 | | FAA STC ST00971SE-D | BOEING 737-300 BOEING 737-400 BOEING 737-500 | Installation of the Rosemount Aerospace Cockpit Door Surveillance system (CDSS). | | 1. Rosemount Aerospace Data List, Doc No. D0530888, Revision B, dated February 15, 2006, or later approved revisions. 2. Rosemount Aerospace Doc No. D0530836, Revision A, dated December 16, 2005, or later approved revisions. 3. Rosemount Aerospace Inst (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 20.04.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|---------------------|----------------|----------------------|---|---|---|--|--|---------------|
| 10015899 | ROSEMOUNT AEROSPACE, Inc. | EASA.IM.A.S.02020 | | FAA STC ST02368CH | BOEING 737-300 BOEING 737-400 BOEING 737-500 BOEING 737-600 BOEING 737-700 BOEING 737-800 BOEING 737-900 | Installation of Rosemount Aerospace Cockpit Door Surveillance System (CDSS) in accordance with Rosemount Aerospace MDL Doc. No. D06400594, Revision B, dated December 29, 2006, or later EASA approved revision. | | - MDL, Doc No. D06400594, Revision B, dated December 29, 2006, or later EASA approved revision - Flight Manual Supplement, Rosemount Aerospace Doc No D06400184, Rev A, FAA approved January 4th, 2007, or later approved revisions. | - Compatibility of this design change with other previously approved modifications must be determined by the installer. - In accordance with Flight Manual Supplement, Rosemount Aerospace Doc No D06400184, Rev A, FAA approved January 4th, 2007, or later approved revisions. | 06.02.2007 |
| 10016410 | ROSEMOUNT AEROSPACE, Inc. | EASA.IM.A.S.02742 | REV. 3 | FAA STC ST02649CH | A319-111/-112/-113/-114/-115 A319-131/-132/-133 A320-111/-211/-212/-214 A320-231/-232/-233 A321-111/-112/-131 A321-211/-212/-213/-231/-232 | Installation of Rosemount Aerospace 8700C2-4 Laptop Docking Station - Electronic Flight Bag (EFB) System Provisions. A319/ A320 / A321 Series 8700C2-4 Laptop Docking Station (LDS) - Electronic Flight Bag (EFB) System Provisions. Revision of the EASA STC (...) | Certification Basis in accordance with EASA Type Certificate Data Sheets A.064. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.1. CRI F (...) | - FAA STC ST02649CH amended July 13, 2009- Rosemount Aerospace Master Data List, ref. doc. No. D06402677, Revision L, dated 9th November 2010- FAA Approved Airplane Flight Manual Supplement (AFMS), ref. Rosemount Aerospace Doc. D06402688, Revision B, (...) | Not applicable | 18.11.2010 |
| 10016424 | ROSEMOUNT AEROSPACE, Inc. | EASA.IM.A.S.02762 | | FAA STC ST02712CH | BOEING 747-400 | Installation of Rosemount Aerospace 8700C2-6 Series Laptop Docking Station (LDS) - Electronic Flight Bag (EFB) Systems Provisions in accordance with Rosemount Aerospace Master Data List, Doc. No. D06402649 Revision C, dated April 3, 2009 or later FAA approved revision. | | Airplane Flight Manual Supplement for B747-400, Doc. No. D06402687 Rev. A. | 1. FAA approved Airplane Flight Manual Supplement, Rosemount Aerospace Doc. No. D06402687, Revision A, approved April 2, 2009 or later FAA approved revision, is required on board of the modified aircraft.2. This installation approval does not i (...) | 23.04.2009 |
| 10016617 | ROSEMOUNT AEROSPACE, Inc. | EASA.IM.A.S.03044 | REV. 2 | FAA STC ST02675CH | A340-211, -212, -213 A340-311, -312, -313 A340-541, -642 | Installation of Rosemount Aerospace 8700C2-3 Laptop Docking Station - Electronic Flight Bag (EFB) System Provisions. Revision of the EASA STC 10016617, Rev. 1 to remove EWIS Limitation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.1. CRI SE-81 "Installed Resources for EFB Class 2 Provisions"2. CRI H-01 "Enhanced Airw (...) | - FAA STC ST02675CH- Rosemount Aerospace Master Data List, ref. doc. No. D06402676, Revision H, dated 5th August 2009- FAA Approved Airplane Flight Manual Supplement (AFMS), ref. Rosemount Aerospace Doc. D06403401, Revision B, dated 7th August 2009- E (...) | This installation approval does not cover the Electronic Flight Bag computers, only the Electronic Flight Bag mounting and wiring. The operational approval of the Electronic Flight Bag computers should follow the guidance in the Temporary Guidance Leaflet TGL 36 | 26.08.2010 |
| 10016655 | ROSEMOUNT AEROSPACE, Inc. | EASA.IM.A.S.03104 | | | BOEING MD 11F | Installation of Rosemount Aerospace 8700C2-8 Series Laptop Docking Station (LDS) - Electronic Flight Bag (EFB) System Provisions in accordance with Rosemount Aerospace Master Data List, Doc. No. D06403165, Revision D, dated April 3, 2009 or later FAA (...) | | Airplane Flight Manual Supplement for MD-11 series Aircraft Rosemount Document No. D06403253 Rev. B | 1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2) FA (...) | 10.06.2009 |
| 10026912 | ROSEMOUNT AEROSPACE, Inc. | P-EASA.IM.A.S.03087 | REV. 1 | FAA STC ST02743CH | B737-300, -700, -800, -900 | Validation of major change to FAA STC concerning EFB Installation of 8700A1-3 Electronic Flight Bag The change to EFB STC adds the following features: implementation of Display Module (DM) ON/OFF backlight connections to WOW and Door Open discretes | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | - Master Document List D06403434 rev G- Airplane Flight Manual Supplement D06403451 rev B or later revisions of the above listed documents approved by EASA | Not Applicable | 01.07.2011 |
| 10027535 | ROSEMOUNT AEROSPACE, Inc. | P-EASA.IM.A.S.03159 | | FAA STC ST02761CH | 737-300, 400, 500 | Installation of Rosemount 8700C2-7 Series Laptop Docking Station (LDS) - Electronic Flight Bag (EFB) System. Validation of FAA STC ST02761CH | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Refer to list of changes on FAA ST02761CH. The certified noise and/or emissions level (...) | Rosemount Aerospace Documents: Master Data List D06403164 Rev. C Approved Airplane Flight Manual Supplement D06403463 Rev. A Instructions for Continued Airworthiness D06403964 Rev. B or later revisions of the above listed documents approved by EASA in (...) | Refer to limitations and conditions on FAA ST02761CH. | 08.10.2009 |
| 10033819 | ROSEMOUNT AEROSPACE, Inc. | P-EASA.IM.A.S.02736 | REV. 1 | FAA STC ST02668CH | 777-200, 777-200LR, 777-300 777-300ER, 777F | Installation of Rosemount Aerospace 8700C2-1 Laptop Docking Station - Electronic Flight Bag (EFB) System Provisions | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Special Condition CRI H-01 EWIS | Rosemount Aerospace Master Data List, Doc. No. D06402489, Revision K, dated 27 December 2011; Airplane Flight Manual Supplement(s) as per Rosemount Aerospace Master Data List, Doc. No. D06402489, Revision K, dated 27 December 2011; Instruction for Contin (...) | This installation approval does not include the Electronic Flight Bag computers. The operational approval of the Electronic Flight Bag computers should follow the EASA Operational Approval process: Any equipment that is connected to the aircraft electr (...) | 18.10.2012 |
| 10041824 | ROSEMOUNT AEROSPACE, Inc. | | | FAA STC ST02944CH | 777-200, 777-200LR, 777-300 777-300ER, 777F | Installation of an AC-DC Converter. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Special Condition CRI H-01 EWIS | Rosemount Aerospace Master Data List, Doc. No. D06408653, Revision B, dated 21 December 2011; Airplane Flight Manual Supplement, Rosemount Aerospace Doc. No. D06408872, Revision A dated 23 December 2011; Instruction for Continued Airworthiness (ICA) / EW (...) | Aircraft modification law Rosemount Aerospace Master Data List, Doc. No. D06408653 Configuration A requires concurrent installation of a system of equipment utilizing the DC power provided by the AC-DC power converter; Aircraft modification law Rosemount (...) | 18.10.2012 |
| 10042935 | ROSEMOUNT AEROSPACE, Inc. | EASA.IM.A.S.01050 | | FAA STC ST02203CH | MD-11 | Installation of Rosemount Aerospace 8400M Series Cockpit Door Surveillance System FAA STC ST02203CH | | CAA-NL report nr. IW/LTB/TCP-06.531095 | 1. FAA approved Airplane Flight Manual Supplement, Rosemount Aerospace CDSS AFM Supplement to McDonnell Douglas MD-11, Document D0431158, Revision A, or later FAA approved revision, is required on board of modified aircraft.2. This STC is approved only (...) | 17.02.2006 |
| 10043061 | ROSEMOUNT AEROSPACE, Inc. | EASA.IM.A.S.03039 | | FAA STC ST01474CH | MD-11, MD-11F | Installation of Rosemount Aerospace Model 1633D Pressure Transducer in accordance with Rosemount Aerospace Master Data List, Document nr. D0130401, Revision C, dated June 1, 2001 or later FAA approved revision. | | None. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The (...) | 04.12.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|-------------------------|----------------|-----------------------|---|---|--|--|--|---------------|
| 10043062 | ROSEMOUNT AEROSPACE, Inc. | EASA.IM.A.S.02965 | REV. 1 | FAA STC ST02667CH | A330 SERIES | Installation of Rosemount Aerospace 8700C2-2 Laptop Docking Station - Electronic Flight Bag (EFB) System Provisions | | - FAA STC ST02667CH Rosemount Aerospace Master Data List, ref. doc. No. D06402672, Revision J, dated 8th June 2009. FAA Approved Airplane Flight Manual Supplement (AFMS), ref. Rosemount Aerospace Doc. 006402689, Revision B, dated 8th June 2009 or ref. R (...) | 1. For this STC the following CRI is applicable and is part of the certification basis:- CRI S-81 Installed Resources for EFB Class 2 Provisions This installation approval does not cover the Electronic Flight Bag computers, only the Electronic Flight (...) | 22.06.2009 |
| 10043265 | ROSEMOUNT AEROSPACE, Inc. | | | FAA STC NO. ST02199CH | 717-200 | Installation of Rosemount Aerospace Inc. 8400E1 Cockpit Door Surveillance System in Boeing 717-200 Series aircraft. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC NO. ST02199CH dated 28 October 2005. Rosemount Aerospace Master Data List, Doc. No. D0530319, Revision B, dated 28 October 2005. Airplane Flight Manual Supplement, Rosemount Aerospace Doc. No. D0530334, Revision A, dated 28 October 2005 or later r (...) | none | 17.01.2013 |
| 10046156 | ROSEMOUNT AEROSPACE, Inc. | | | FAA STC NO. ST02819CH | A330-200/-300 | Smart Display EFB System. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC NO. ST02819CH dated 21 March 2012. Rosemount Aerospace Master Data List, Doc. No. D06404080, Revision Y, dated 15 August 2013. Airplane Flight Manual Supplement as identified in Rosemount Aerospace Master Data List, Doc. No. D06404080, Revision Y, dated 15 August 2013 or later revisions of the a | For Rosemount Aerospace Master Data List, Doc. No. D06404080, Configurations A and C (Class 3 EFB), the Rosemount Aerospace Electronic Flight Bag (EFB) system is an approved Class 3 EFB system for use in existing Type A and Type B software applications f (...) | 22.08.2013 |
| 10016035 | ROSEN SUNVISOR SYSTEMS L.L.C. | EASA.IM.A.S.02204 | REV. 2 | FAA STC SA4147NM | BAE 125 SERIES 1000A BAE 125 SERIES 800A HAWKER 1000 HAWKER 800 XP HAWKER 850 XP HAWKER 900 XP | Installation of Cockpit Sun Visors. | | In accordance with Rosen drawing List Number RBA-800-00DL, Revision B, dated May 10, 2007, or later approved revisions. | | 21.05.2008 |
| 10016661 | ROSEN SUNVISOR SYSTEMS L.L.C. | EASA.IM.A.S.03124 | | FAA STC SA01838SE | | Installation of Sunvisor. | | According to Rosen Sunvisor System, LLC Master Drawing List No. 1381000-00DL, Rev. B or later FAA approved Revisions. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 23.04.2009 |
| 10026892 | ROSEN SUNVISOR SYSTEMS L.L.C. | P- EASA.IM.A.S.03271 | | | FAN JET FALCON C, D, E, F, G | EASA Validation of FAA STC SA2652NM Cockpit Sun Visor Installation i.a.w. Rosen Drawing List No. RF 20-00DL, Revision D dated 14th November 2008 or later revisions approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004/04/CF (or subsequent revisions to these decisions) | Original EASA Certification Basis for Fan Jet Falcon and Fan Jet Falcon C, D, E, F and G as per French DGAC TCDS's Number 103 Issue 13 dated April 1981 and TCDS's Number 103 bis Issue 6, dated October 1986. | See those of original FAA STC SA2652NM | None | 19.08.2009 |
| 10027193 | ROSEN SUNVISOR SYSTEMS L.L.C. | P- EASA.IM.A.S.03219 | | FAA STC SA2614NM | 1. MYSTERE FALCON 200 2. MYSTERE FALCON 50 | EASA Validation of FAA STC SA2614NM Cockpit Sun Visor Installation in accordance with Rosen Drawing List No RF 50/200-00DL, Revision C dated 26th August 2009 or later revisions approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | See those of original FAA STC SA2614NM or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | This EASA approval is also applicable to the Falcon 50EX version of the MF50 aircraft model as per this aircraft version definition described on EASA TCDS.A.062 | 15.09.2009 |
| 10031251 | ROSEN SUNVISOR SYSTEMS L.L.C. | P- EASA.IM.A.S.03270 | | FAA STC SA00871SE | SEE EASA APPROVED MODEL LIST | Installation of Sunvisor | CS 23.C, CAR 23 | According to Rosen Sunvisor System, LLC Master Drawing List No. 9050-0118-013, Rev. F or later FAA approved Revisions | None | 04.08.2010 |
| 10033410 | ROSEN SUNVISOR SYSTEMS L.L.C. | | | FAA STC SA3529NM | DHC-6-100, -200, -300 | Cockpit Sun Visor installation in accordance with FAA STC SA3529NM | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rosen Drawing List Number RTO-00DL, Rev. A, dated June 29, 2010. Rosen Sunvisor Systems Installation Instruction, Doc. 9041-0169-001, Rev. A, dated June 29, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED (...) | none | 19.01.2011 |
| 10039248 | ROSEN SUNVISOR SYSTEMS L.L.C. | | | FAA STC SA2367NM | CL-600-1A11/-2A12/-2B16/-2B19 CL-600-2C10/-2D24 | Cockpit Sunvisor Installation | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. SA2367NM dated 04 June 2004. Rosen Sunvisor Systems, Master Drawing List No. RCL 6000501-00DL Revision E, dated 29 June 2010, or later approved revision or later revisions of the above listed documents approved by EASA in accordance with EA (...) | None | 20.04.2012 |
| 10045675 | ROSEN SUNVISOR SYSTEMS L.L.C. | | | FAA STC SA2150NM | CESSNA 500, 550, S550, 560 | Rosen Sunvisor System Monorail mounted cockpit sunvisor | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rosen Sunvisor Systems, Drawing List R500S-00DL, revision G, dated 22 January 2009. Rosen Sunvisor Systems, Installation Instructions for Cessna Citation Monorail Sunvisor System Kit R500S-010-1, Doc. 9041-0126-001, rev D, date 21 January 2009. Rosen Sunv (...) | None | 11.07.2013 |
| 10028673 | ROSS FRESNO LLB | EASA.IM.A.S.03153 | | | CESSNA 441 SERIES | | | | | 15.07.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|------------------|----------------|-------------------------------|-----------------------------|--|---|---|--|---------------|
| 10032879 | ROTORCRAFT SUPPORT, Inc. | | | FAA STC SR01456LA | 407 | Installation of Concorde RG407 Battery | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | RS407-010 revision C dated 30 June 2003 - Master Document List; RS407-006 revision NC dated 20 May 2003 - Installation Instructions; RS407-007 revision NC dated 30 June 2003 - Flight Manual Supplement; 5-0144 revision E dated 19 October 2004 - Concorde (...) | None | 06.12.2010 |
| 10017109 | ROTORTECH LIMITED | EASA.R.S.01085 | | | EUROCOPTER FRANCE EC 155 B1 | Compendium Modification. Compilation of 22 individual modifications associated with the installation of equipment and systems enabling the transfer of EC 155B1 S/N 6642 from FAA register (N84AZ) to the UK register (G-NIVA). | | Rotortech Assignment Report P446/155/MO dated June 9 2005 or later approved revision; Rotortech RFM Supplement; EC155B1/01 Issue 1 WX-500 Stormscope Weather Mapping Sensor; EC155B1/02 Issue 1 UNS UL-701 Unlink Communication Management System; EC155B1/03 (...) | 1. This STC is applicable to EC155B1 S/N 6642 only. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.09.2005 |
| 10033185 | ROW 44, Inc. | | REV. 1 | FAA STC ST02829CH | B737-700/-800/-900 | Installation of Row 44 Broadband Antenna System Provisions | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-25 Amdt 1 (per effective date 12 Dec 2006 would be equivalent to FAA Cert Basis FAR- (...) | Master Data List, Doc. No. DC104-7023-00, Revision C, dated April 13, 2010 - Instructions for Continued Airworthiness as identified in Master Data List, Doc. No. DC104-7023-00, Revision C, dated April 13, 2010 or later revisions of the above listed document (...) | The equipment for which these provisions are intended has not been certified. Additional approval is required for the installation of this equipment and must be evaluated to assure satisfactory compliance with the applicable airworthiness standards. | 22.10.2012 |
| 10033186 | ROW 44, Inc. | | REV. 2 | FAA STC ST02836CH-D | B737-700, -800, -900 | Installation and activation of a Row 44 Broadband System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-25 Amdt 1-CRIF-01 "Installation of Wireless Local Area Networks (WLAN)" <-> CRIF H (...) | Master Data List, Doc. No. DC104-7024-00, Revision G, dated January 31, 2011 - Airplane Flight Manual Supplement in accordance with Armstrong Aerospace Master Data List, Document No. DC104-7024-00, Revision G, dated January 31, 2011 - Instructions for Conti (...) | The Row 44 Wireless Broadband System is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other use of this equipment will require a re-examination of the certification base (...) | 08.11.2012 |
| 10044038 | ROW 44, Inc. | | | FAA STC ST2904CH | B757-200 | Installation of the Row 44 Wireless Broadband System Provisions (Description: Reference Envoy Aerospace Project Specific Certification Plan, Document No. E10010-2500-C001. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: 25.29(a), (b) Amdt. 25-72 - equivalent of CS-25 Amdt. 225.251 Amdt. 25-77 - equivalent of CS-25 Amdt. 225.605(a) Amdt. 25- (...) | Master Drawing List, Row 44 document No. SIE-346-703, Revision L, dated February 12, 2013 - Instructions for Continued Airworthiness, Row 44 document No. SIE-407-607 <-> Revision A, dated February 13, 2013 or later revisions of the above listed document (...) | The aircraft must be maintained in accordance with Section 2 - Airworthiness Limitations of Instructions for Continued Airworthiness, Row 44 document No. SIE-407-607, Revision A, dated February 13, 2013 Prior to installation of this design change it must (...) | 13.03.2013 |
| 10046518 | ROW 44, Inc. | | | ST02258LA-D | 757-200 SERIES | Installation of the ROW 44 satellite-based Wi-Fi broadband internet connectivity system. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: 25.29(a), (b) Amdt. 25-72 - equivalent of CS-25 Amdt. 225.251 Amdt. 25-77 - (...) | Master Data List, Row 44 document No. SIE-346-701, Revision G, dated September 9, 2013 - Airplane Flight Manual Supplement, Row 44 document No. SIE-346-604, Revision A, dated 6 September, 2013 - Instructions for Continued Airworthiness, Row 44 document No. (...) | None | 27.09.2013 |
| 10016327 | ROYAL ATLANTIC AVIATION, INC. | EASA.M.A.S.02636 | | FAA STC SA02636AT | CESSNA 441 SERIES | Ground Recognition Strobe Light System | | Installation law. Master Drawing List MDL RAA-MDL-1, Rev. A, or later FAA approved Revisions and RAA-INST-1, Rev. B, or later approved Revisions. Document No RAA-ICA-1, Rev. IR, or later approved Revisions | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the aircraft (...) | 22.09.2008 |
| 10017127 | RTE-STH | EASA.R.S.01215 | REV. 1 | | AS 355N | New pilot/co-pilot manual quick release pyrotechnic sub-system, P/N JD3448AB00100-A0, for approved HEC devices ref. 01-STH-001 and 01-STH-002 and AAN 24527 | | Refer to RTE-STH technical file ref. 2005-STH-001&002, ind. 2. Definition: based on PyroAlliance design file RDD 0 JD3448AB00100 01 Installation, Inspection and Maintenance in accordance with PyroAlliance document STU 0 JD3448AB00100, evolution A, Rev 1 Op (...) | 1. Limited to the following AS 355 N aircrafts: S/N 5649, F-GSTH equipped with DGAC-F approved design change ref. 01-STH-001 or 01-STH-002-S/N 5543, F-GJAX equipped with DGAC-F approved design change ref. 01-STH-001-S/N 5566, G-LINE equipped with C (...) | 08.08.2006 |
| 10017135 | RTE-STH | EASA.R.S.01229 | | | AS 332 C1 | Installation for Human External Cargo (HEC) applications | | Definition: RTE-STH Master Drawing List, report no DT/ETUD/CNER-STH-PAMM/05/01270, issue 1 Installation: RTE-STH "Instructions Techniques de Montage", report no DT/ETUD/CNER-STH-PAMM/05/01263, issue 1 Inspection and Maintenance: RTE-STH "Manuel de Mai (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 21.12.2005 |
| 10017165 | RTE-STH | EASA.R.S.01273 | | DGAC-F TC N° 168 | AS 355 N | Personnel transport aboard hooked baskets as rotorcraft-load combinations for human external cargo (HEC). | | Definition: according to Master Drawing List DT-IMR-CNER-STH-PAMM-06-01059 ind.3, dated 12/07/07-Installation: according to ITM document (Instructions Techniques de Montage) GUI-IMR-CNER-STH-PAMM-07-00535 ind.1, dated 24/08/07-Inspection and Maintenance (...) | This STC is also applicable on AS355N when equipped with EASA.R.S.01366. | 18.10.2007 |
| 10017210 | RTE-STH | EASA.R.S.01366 | | DGAC FRANCE STC FAM 98STH0595 | AS 355 N | Helicopter equipped with ARRIUS 1A modified named ARRIUS 1E. DGAC-F approved design change 98STH0595 relates to installation of two modified ARRIUS 1A engines on the AS355N helicopter. This installation allows increased engine performance required to p (...) | | # Manuel d'installation ARRIUS 1E : X319 HS 0011# Supplement au manuel de vol: DT-ETUD-CNER-STH-PAMM-07-00315 rev.1# Additif au Manuel d'entretien Turbomoteur GBF 124# Original DGAC approval FAM 98 5TH 0595 | # Engine operating limitations as defined in the #Installation Manual# reference X319 HS 0011.# Engine maintenance requirements defined in Manuel d'entretien Turbomoteur GBF124. | 23.03.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|----------------|----------------|----------------------|--|---|---|--|---|---------------|
| 10039175 | RUAG AEROSPACE | EASA.A.S.00200 | REV. 2 | | DHC-6-300 | Minor change to STC EASA.A.S.00200 by adding the option to install aThommen AD32 instead of the Shadin ADC 2000Installation of TAWS class A Honeywell MK VI incl. KMD-850 indicator in accordance with Certification Program Doc. T6-100 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate approval.FAR 23The certified noise and/or emissions levels of the original product are unchanged (...) | -Master Document List, Doc. No. T6-000, Rev. 000, dated 15.05.2006-AFMS Doc. No. T6-500, Rev. 1, dated 15.05.2006or later revisions of the above listed documents approved by EASA | None | 16.04.2012 |
| 10013955 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.00505 | | | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 (CL-601) | Upgrade Dual Honeywell FMS NZ-2000-PRNAV. | | - Master Document List 100A-MDL, Airplane Flight Manual Supplement 100A-AFMS, Continued Airworthiness Plan 100-CAWP | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 12.01.2005 |
| 10013956 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.00506 | | | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 (CL-601) | Mode-S Transmitter Collins TDR 94D upgrade - ELS/EHS, Associated Technical Documentation: Master Document List 100C-MDL, Airplane Flight Manual Supplement 100C-AFMS, Continued Airworthiness Plan 100C-CAWP | | - Master Document List 100C-MDL, Airplane Flight Manual Supplement 100C-AFMS, Continued Airworthiness Plan 100C-CAWP | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 12.01.2005 |
| 10013980 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01008 | | | CESSNA 525 CESSNA 525A | Installation of Honeywell EGPWS MK V with RAAS | | Doc. No. 128B-MDL, 128B-EOEASA approved RUAG AFM-S Doc. No. 128B-AFMS, Original issue, or later approved RevisionRUAG Continued AW procedures. Doc. No. 128B-CAWP, Original issue, or later approved Revision | | 10.04.2007 |
| 10013981 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01012 | | | BOMBARDIER CL-600-2B16 BOMBARDIER CL-600-2B16 (CL-601) | Installation of Honeywell EGPWS MK V. | | - Master Document List 100B-MDL, rev. N/C dated 06.04.2005 or later EASA-approved Revisions.- Airplane Flight Manual Supplement 100B-AFMS, rev. N/C dated 06.04.2005 or laterEASA-approved Revisions.- Continued Airworthiness Plan AMM. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 12.04.2005 |
| 10014020 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01075 | | | CESSNA 560 | Installation of an ACAS II (TCAS 2000, CH 7) with Dual Mode S Transponders ACSS RCZ-852-ELS/EHS. | | - Design Modification Document Doc. No. 039A-MDL.- Operating Instructions Doc. No. 039A-AFMS.- Instructions for Continued Airworthiness Doc. No. 039A-CAWP. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 27.01.2005 |
| 10014021 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01076 | | | CESSNA 501 | Installation of Dual Mode S Transponders ACSS RCZ-852-EHS | | -Airplane Flight Manual Supplement Doc.-No. 096B-AFMS, Rev. N/Cor later EASA approved revision | | 01.03.2005 |
| 10014034 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01107 | | | CESSNA 550 | Installation of TCAS 2000, CH 7 with Dual Mode S transponders ACSS RCZ-852-EHS-Master Document List, Doc No 147A-MDL | | -Master Document List, Doc No 147A-MDL-Airplane Flight Manual Supplement, Doc No 147A-AFMS-Continued Airworthiness PlanDoc No 147A-CAWP-LBA TCDS No 2048, Issue 1 | | 31.01.2005 |
| 10014035 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01108 | | | CESSNA 550 BRAVO | Upgrade of GPWS to EGPWS | | - AFM Supplement 5387/04/1-06 | | 10.02.2005 |
| 10014044 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01122 | | | CESSNA 560 | Installation of ACAS II CAS 67A with Dual Mode S Transponder RCZ-833 J-ELS | | -System Description Doc No 085F-MDL-Airplane Flight Manual supplement Doc No 085-AFMS-Instructions for Continued Airworthiness Doc No 085F-CAWP | | 17.03.2005 |
| 10014060 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01144 | | | CESSNA 560 (CITATION V) | Installation of ACAS II (ACSS TCAS 2000, CH7) with Dual Mode S Transponders RCZ-833J ELS | | -Master Document List, Dok Nr. 093C-MDL, Rev N/C dated 03.03.04-AFM Supplement, Dok Nr. 093C-AFMS, Rev N/C dated 03.03.04-Continued Airworthiness Plan (Maintenance Manual Supplement), Dok Nr. 093C-CAWP, Rev N/C dated 03.03.04 | | 10.03.2005 |
| 10014085 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01176 | REV. 1 | | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 (601-3A) BOMBARDIER CL-600-2B16 (601-3R) | Upgrade of the two existing Mode-S Transponders ACSS RCZ-852 by ACSS Service Bulletin 8002246-001 for ELS/EHS.Replacement of the Controller, Collins CTL-92 by Mode S Controller, Collins CTL-92E.ACSS Mode-S transponder RCZ-852 ELS/EHS with Collins CTL-92E. | | -Master Document List, Doc No 043A-MDL, Rev. N/C or later EASA approved Revisions-Airplane Flight Manual Supplement Document No 043A-AFMS, Rev. A or later EASA approved Revisions-RUAG Master Document List No. 191C-MDL Revision A or later approved rev (...) | | 29.02.2008 |
| 10014161 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01293 | | | BOEING DC-9-82 (MD-82) BOEING DC-9-83 (MD-83) | Installation of Mode S Transponder ACSS XS-950 ELS/EHS with Control Gables G7490-66 including associated Airplane Flight Manual Supplement | | RUAG Master Document List Doc No 042A-MDLRUAG Airplane Flight Manual Supplement Doc No 042A-AFMS, Rev N/C dated 1 June 2005 | | 28.06.2005 |
| 10014189 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01331 | | | CESSNA 525 | Installation of ACAS II System with Dual Mode S Transponder. ELS compliant and EHS capable | | AFM Supplement Doc. No. 118A-AFMS, revision 0 or later EASAapproved revision. | | 18.07.2005 |
| 10014195 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01337 | | | BOMBARDIER BD-700-1A10 | Installation of MicroQAR acc.046A-SD | | | | 21.06.2005 |
| 10014204 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01355 | | | BOMBARDIER DHC-8-400 | Installation of YDL 4000/A VHF Data Link Mode 4 (CNS Systems) and installation of a combined VHF/GPS Antenna | | Master Document List Doc. No. 045A-MDL, Rev. N/C or later EASA approved revisions. Continued Airworthiness Plan Doc. No.045A CAWP | | 18.07.2005 |
| 10014205 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01356 | | | CESSNA 500, CITATION | Installation of a TWS/RMI Sandel ST 3400 (TAWS Class B) | | - Master Document List Doc. No. 152A-MDL-Airplane Flight Manual Supplement Doc. 152A-AFMS | | 24.08.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|----------------|----------------|----------------------|--|---|---|--|--|---------------|
| 10014206 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01357 | | | CESSNA 500, CITATION | Installation of Dual Mode - S Transponder System Garmin GTX-330D - Elementary Surveillance. (Removal of the existing Rockwell Collins TDR-90 transponder). Installation of two additional L-Band Antennas on top of the fuselage. | | - Master Document List Doc. No. 152B-MDL-Airplane Flight Manual Supplement Doc. 152B-AFMS | | 24.08.2005 |
| 10014377 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01652 | | | CESSNA 560 | Installation of a Honeywell Solid-State Flight Data Recorder (SSFDR) with Teledyne Controls Mini Flight Data Acquisition Unit (MFDAU) | | Engineering Order 148A EO, dated 02.01.2006Airplane Flight Manual Supplement, Doc. 148A-AFMS, Issue N/C, dated 09.01.2006Maintenance Manual Supplement, Doc. 148A-AMMS, dated 09.01.2006 | | 20.01.2006 |
| 10014407 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01702 | | | CESSNA 550 CITATION II | Honeywell EGPWS Mk VIII | | Change Notice No: 170B-CN Rev. N/C or later approved RevisionAirplane Flight Manual Supplement 170B-AFMS Rev. N/C or later approved Revision | Certification Basis of the Change: FAR 25 as defined in TCDS A22CE, Rev. 58, additionally JAA Leaflet No. 12 (TAWS) | 24.04.2006 |
| 10014409 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.01705 | REV. 1 | | CESSNA 650 | Upgrade of two existing Mode S transponders, Rockwell Collins TDR-94D by Rockwell Collins SB 501. Replacement of Mode S Controller CTL-92 by CTL-92E along with installation of new wiring for Enhances Surveillance Revision 1 to include EASA AD 2009-0173, (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Document List 163A-MDL, Rev. A from 27 October 2010Airplane Flight Manual Supplement 163A-AFMS, Rev. A from 26 October 2010 or later revisions of the above listed documents approved by EASA | Aircraft Serial Numbers 650-7001 through 650-7119 | 29.10.2010 |
| 10014429 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02475 | | | CESSNA 550 CITATION II | ACAS II 5CAS67A) with Dual Mode S XPDR MST-67A-EHS | | Change Notice No: 170A-CN Rev. N/C or later approved Revision Airplane Flight Manual Supplement 170-AAFMS Rev. N/C or later approved Revision | Certification Basis of the Change: FAR 25 as defined in TCDS A22CE; Rev. 58, additionally JAA Leaflet No. 8, Rev. 2 (ACAS II), JAA Leaflet No. 13, Rev. 1 (Mode-S ELS) and JAA NPA 20-12a (Mode-S EHS) | 19.04.2006 |
| 10014571 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02675 | | | | Installation of Honeywell Solid-State Flight Data Recorder (SSFDR) with Teledyne Controls Mini Flight Data Acquisition Unit (MFDAU) | | Engineering Order, 036A-EO, dated 06/06/2006 or later approved revisionAFM Supplement, 036A-AFMS, dated 10/05/2006 or later approved revision | -- | 26.06.2006 |
| 10014599 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02720 | | | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 (601-3A) BOMBARDIER CL-600-2B16 (601-3R) BOMBARDIER CL-600-2B16 (604) | Enhanced Ground Proximity Warning System-EGPWS-MK VII or MKV enhancement with RAAS | | RUAG Aerospace Services Modification reference SE-464b/2006/w/069B-RFCRUAG Aerospace AFM Supplement 069B-AFMS Revision N/C dated November 09 2006 or later approved revision | | 07.02.2007 |
| 10014610 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02737 | | | CESSNA 525 | Installation of KLN-90B Model: 525 S/N 0001 up to 0359 | | Engineering Order 118-EORUAG AFM-S 118-FM, Rev. 0, approved 10. March 2004, or later EASA approved RevisionMaintenance Procedures 118-MDL, Rev. 0, or later approved Revision | S/N 525-0001 up to 525-0359 This STC is approved only for the product configuration as defined in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 10.07.2006 |
| 10014623 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02757 | | | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 (CL-601) | Honeywell Primus II RCZ-833 () Integrated Radio System # EHS. | | RUAG Aerospace Services Master Drawing List 179E-MDL. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and #Associated Technical Documentation#. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 12.01.2007 |
| 10014624 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02758 | | | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 (CL-601) | Upgrade Dual Honeywell FMS NZ-2000 / PRNAV. | | RUAG Aerospace Services Master Drawing List 179B-MDL. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and #Associated Technical Documentation#. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 12.01.2007 |
| 10014625 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02759 | | | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 (601-3A) | Indium Satcom ST3100. | | RUAG Aerospace Services Master Drawing List 179C-MDL. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and #Associated Technical Documentation#. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 12.01.2007 |
| 10014645 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02799 | | | BOMBARDIER CL-600-2B16 (CL-601) | 40Ah Battery with Charger. | | RUAG Aerospace GmbH Master Drawing List 179F-MDL. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and #Associated Technical Documentation#. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 12.01.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|----------------|----------------|----------------------|--|--|---------------------|---|---|---------------|
| 10014646 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02800 | | | BOMBARDIER CL-600-2B16 (601-3A) | Interior, Aircell Antenna & Miscellaneous Equipment. | | RUAG Aerospace Services Master Drawing List 179G-MDL. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "#Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 12.01.2007 |
| 10014648 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02803 | | | CESSNA 560 (CITATION V) | Transponder Mode S Honeywell Primus II # ELS/EHS, Upgrade of the two existing Mode S Transponder RCZ-833K by Honeywell Service Bulletin to implement ELS/EHS transponder capability. Installation of new wires from the FMS and the two Display/Guidance C (...) | | RUAG Compliance Check List Doc.-No.:148C-CCL Rev N/C dated 24.08.06, RUAG System Description Doc.-No.:148C-SD Rev. N/C dated 18.08.PATS Aircraft, LLCRUAG AFM Supplement Doc.-No 148C-AFMS | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 15.09.2006 |
| 10014654 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02811 | | | CESSNA 650 CITATION III CESSNA 650 CITATION VI CESSNA 650 CITATION VII | Installation of Flight Information System Sirius FD200 | | Installation in accordance with RUAG Master Document List Document No 178D-MDL and Engineering Order Document No 178D-EO, at the latest approved revision. | None | 11.12.2006 |
| 10014655 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02812 | | | CESSNA 650 CITATION III CESSNA 650 CITATION VI CESSNA 650 CITATION VII | Installation of Honeywell EGPWS Mk VIII. | | Installation in accordance with RUAG Master Document List Document No 178B-MDL and Engineering Order Document No 178B-EO, at the latest approved revision.Aircraft to be Operated in accordance with RUAG Aircraft Flight Manual Supplement Document No 178B-AFMS, at the latest approved revision. | None | 11.12.2006 |
| 10014656 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02813 | | | CESSNA 650 CITATION III CESSNA 650 CITATION VI CESSNA 650 CITATION VII | Installation of Honeywell MST-67A Mode S Transponders. | | Installation in accordance with RUAG Master Document List Document No 178A-MDL and Engineering Order Document No 178A-EO, at the latest approved revision.Aircraft to be Operated in accordance with RUAG Aircraft Flight Manual Supplement Document No 178A-AFMS, at the latest approved revision. | None | 11.12.2006 |
| 10014657 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02814 | | | CESSNA 650 CITATION III CESSNA 650 CITATION VI CESSNA 650 CITATION VII | Installation of Aircell Iridium Satcom Access II | | Installation in accordance with RUAG Master Document List Document No 178E-MDL and Engineering Order Document No 178E-EO, at the latest approved revision.Aircraft to be Operated in accordance with RUAG Aircraft Flight Manual Supplement Document No 1 (...) | None | 11.12.2006 |
| 10014661 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02818 | | | CESSNA 650 CITATION III CESSNA 650 CITATION VI CESSNA 650 CITATION VII | Installation of modified and refurbished cabin interior | | Ruag Master Document List Document No. 178F-MDL Engineering Order Document No. 178F-EO, at the latest approved revision. | | 12.12.2006 |
| 10014766 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02958 | | | CESSNA CITATION 501 | Installation of Honeywell Integrated Hazard Avoidance System (HAS) | | Master Doc. List: 162B-MDL, Engineering Order: 162B-EOEASA approved RUAG AFM-S Doc. No. 162B-AFMS, Rev. A, or later approved RevisionRUAG MM-S Doc. No. 162B-AMMS, Original issue, or later approved Revision | This installation does not comply to JAR OPS 1 operational requirements (Class B equipment), thus it is up to installer to comply with national commercial operational rules and requirements. | 25.06.2007 |
| 10014790 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02991 | | | CESSNA 650 CITATION III CESSNA 650 CITATION VI CESSNA 650 CITATION VII | Upgrade of two Mode-S transponders Honeywell MST-67A by Honeywell SB to implement ELS/EHS transponder capability.ATC Controller KFS-578A is exchanged by Mode S ControllerPS-578A.New wires from the transponders to the FMS. | | Document 185A-RFC | According to EUROCONTROL definition this equipment with approval EASA.A.S.02991 is non-EHS compliant. | 05.04.2007 |
| 10014791 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02992 | | | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 (601-3A) BOMBARDIER CL-600-2B16 (601-3R) | Honeywell MST-67A Transponder Mode-S ELS/EHS | | RUAG Aerospace Services Master Drawing List 189A-MDL or later approved revision.RUAG Aerospace Services AFMS 189A-AFMS or later approved revision. | | 04.04.2007 |
| 10014792 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.02994 | | | BOMBARDIER CL-600-2A12 (CL-601) | Inflight Entertainment System (IFE) Sirius | | RUAG Master Document List No. 189B-MDL, Rev. A | | 29.05.2007 |
| 10014819 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03029 | | | BOMBARDIER CL-600-2B16 (CL-604) | EASA Recertification of Cabin Modifications. | | RUAG Master Document List No. 145G-MDL | | 18.10.2007 |
| 10014860 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03084 | | | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 (CL-601) | Honeywell EGPWS MK VII | | Modification: 076E-CNAFM: 076E-AFMS Rev N/C dated 16/04/07 or later approved revisionMM: 076E-EO | | 11.05.2007 |
| 10014861 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03085 | | | BOMBARDIER CL-600-1A11 (CL-600) | ACSS Mode-S Transponder RCZ-852 - ELS/EHS | | Modification: 076F-CNAFM: 076F-AFMS Rev N/C dated 19/04/2007 or later approved revisionMM: 076F-EO | With this modification now embodied the aircraft is non EHScompliant. | 11.05.2007 |
| 10014866 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03090 | | | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) | Dual TAWS Sandel ST3400 - Class A | | RUAG Master Document List No. 191D-MDL Revision A or later approved revision.RUAG Airplane Flight Manual Supplement No.191D-AFMS Revision A or later approved revision. | | 29.02.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|----------------|----------------|----------------------|--|--|---------------------|--|--|---------------|
| 10014880 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03108 | | | CESSNA CITATION 501 CESSNA CITATION CS 551 | Installation of Dual Mode S XPDR MST-67A.Model: 501 S/N 0001 and on (Citation I-SP)551 S/N 0001 and on (Citation II-SP) | | # Master Doc. List: 204A-MDL, Engineering Order: 204A-EO.# EASA approved RUAG AFM-S Doc. No. 204A-AFMS, Original Issue, or later approved Revision.# RUAG MM-S Doc. No. 204A-AMMS, Original Issue, or later approved Revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 21.06.2007 |
| 10014889 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03119 | | | DORNIER 328-300 | Modification of rear-facing seat for use during T/O and Landing.Backrest support of the already certified VIP seats AMP2524...xxx is modified. The seats are now certified for T/O and landing in aft facing condition with a backrest support.ref. TSO-C127 dynamic tests | | Compliance Statement, Doc-Nr.: 155F-CSStress Report U155F530001Airplane Flight Manual Supplement, Doc-Nr.: 155F-AFMS | | 19.11.2007 |
| 10014890 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03120 | | | DORNIER 328-300 | EASA Certification of Installed of Galley and Lavatory Water System.Certified earlier by AvCraft USA,Ref. FAA STC nr. 9993. | | Compliance Statement, Doc-Nr.: 155G-CSMaster Document List, Doc-Nr.: 155G-MDLAirplane Flight Manual Supplement, Doc-Nr.: 155G-AFMS | | 13.11.2007 |
| 10014891 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03121 | | | BOMBARDIER CL-600-2A12 (CL-601) | EASA recertification of Cabin Modifications | | RUAG Master Document List No. 191A-MDL Revision A or later approved revision. | | 29.02.2008 |
| 10014892 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03122 | | | BOMBARDIER CL-600-2A12 (CL-601) | EASA recertification of Avionics Modification | | RUAG Master Document List No. 191B-MDL Initial Issue or later approved revision.RUAG Airplane Flight Manual Supplements.No. 191B-AFMS-TCASNo. 191B-AFMS-FMSNo. 191B-AFMS-COMMNo. 191B-AFMS-HFNo. 191B-AFMS-CVRNo. 191B-AFMS-DFDRat initial issue or later approved revision. | | 29.02.2008 |
| 10014895 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03125 | | | DORNIER 228-212 | Adjustment of Special Fatigue Life.Model: Dornier 228-212, S/N 8181 and 8183 | | Supplemental Time Limits/Maintenance Checks Manual for aircraft S/N 8181TM-STLMCM-8181-090707 Issue July 09/2007Supplemental Time Limits/Maintenance Checks Manual for aircraft S/N8183TM-STLMCM-8183-090707 Issue July 09/2007. | 1. This STC is only applicable to aircraft S/N: 8181, 8183.2. Prior to installation of this modification the installer must determine that the interrelationship between thismodification and any other previously installed modification will intr (...) | 08.10.2007 |
| 10014935 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03190 | | | BOMBARDIER CL-600-2B16 (601-3A) | Installation of a Honeywell Bendix/King Radio KTR 909 System. | | RUAG Aerospace Services Modifications Reference SE4-623b/2007-mo (078E-RFC).RUAG Aerospace Services Master Document List No. 078E-MDL.RUAG Aerospace Services AFM Supplement No. 078E-AFMS. | Restricted to ASN 5105. | 10.10.2007 |
| 10014938 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03195 | | | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-601-3A | Aircell Iridium Satcom, Aaccess EZ (Magnaster C-2000 replacement). | | RUAG Master Document List No. 209B-MDL | | 21.08.2007 |
| 10014954 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03219 | | | CESSNA 650 CITATION III CESSNA 650 CITATION VI CESSNA 650 CITATION VII | Honeywell EGPWS Mk VIIDescription: replacement of the existing GPWS Mk VII with a Honeywell EGPWS Mk VII. Terrain presentation on the EFIS MFD.Installation of indicators/switches in the instrument Panel. | | Installation in accordance RUAG Master Document List Doc-No. 163C-MDL. Engineering Order Doc-No. 163C-EO, latest approved revisions.To be operated in accordance with RUAG AirplaneFlight Manual Supplement Doc-No. 163C-AFMS, latest approvedrevision. | | 08.10.2007 |
| 10014964 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03236 | | | BEECH F90 | Avionics Installation, consisting of- Dual GNS430A VHF Com/GPS and NAV- Dual GAD42 Adapter for GNS430/RMI- Dual KMR675 Marker Beacon Receiver- EHS GTX330D Diversity Transponder and Transponder Top Antenna- GMX2300 Traffic/Radar and JeppiewMFD Displa (...) | | Installation in accordance with RUAG Master Document List Doc No. 210A MDL, Engineering Order Doc No. 210A EO, latest approved revisions.To be operated in accordance with RUAG Airplane Flight Manual Supplement Doc No. 210A AFMS, latest approved revision. | None. | 12.12.2007 |
| 10014995 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03292 | | | | Installation of dual Garmin GBS530AW NAV/COMM/GPS and dual sandel SN4500 EHS. | | POH supplement AS9412, Edition 1Engineering Order: EO-228-234, Rev. 0Continued Airworthiness: TM-TLMCM-090305-ALLChapter 05-21-10 | Vertical approach navigation basing on GPS/WAAS is o nly approved in VMC.Restricted to aircraft equipped with HoneywellKing Silver Crown Series. | 17.03.2008 |
| 10015005 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03305 | | | CESSNA 650 CITATION III CESSNA 650 CITATION VI CESSNA 650 CITATION VII | FMS Upgrade to Dual Honeywell NZ-2000/PRNAV | | Aircraft Flight Manual Supplement, Doc. 223A-AFMS, Rev. IR or later approved revision.Continued Airworthiness, Doc. 223A-CAWP, Rev. IR or later approved revision. | none | 16.01.2008 |
| 10015041 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03357 | | | CESSNA 525 CESSNA 525A | Iridium Satcom AirCell ST3100.Description: Replacement of the previously installed AirCell AT.01 Air Telephone System by AirCell ST 3100 satellite Communication System (SATCOM). | | 1. Installation in accordance with RUAG Master Document List Doc-No. 225B-MDL, Engineering Order Doc-No. 225B-EO, latest approved Revision;2. To be maintained in accordance with RUAG Airplane Maintenance Manual Supplement 225B-AMMS, or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 08.02.2008 |
| 10015054 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03375 | | | DORNIER DO 228-202K | Increase of Fatigue Life Limit. | | Supplemental Time Limits/Maintenance Checks Manual for Aircraft S/N 8046, TM-STLMCM-8046-180405 Rev. 1 | This STC is only applicable to Aircraft S/N: 8046. | 19.12.2007 |
| 10015094 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03431 | | | BEECH B200 | Dual Collins TDR-94() Upgrade-ELS/EHS and Antenna Diversity.Upgrade of existing Dual Collins TDR-94() Transponders by TDR-94D and Replacement of Control CTL-92 by CTL-92E.Installation of two additional top-mounted L-Band Antennas for antenna diversi (...) | | Installation in accordance with RUAG Airplane Flight ManualSupplement Doc-No. 231A-B200-34.80.01AFMS, latest approved revision. | None | 08.02.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|------------------|----------------|----------------------|--|--|--|---|--|---------------|
| 10015124 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03474 | REV. 1 | | FALCON 20 | Dual Universal UNS-1L FMS Installation. | | Master Document List 236A-MDL Revision Original, Engineering Order 236A-F20-34.80.02EO Revision Original, Airplane Flight Manual Supplement 236A-F20-34.80.02AFMS Revision N/C, Maintenance Manual Supplement 236A-F20-34.80.02MMS Revision N/CO later approved revisions of these documents. | This STC is limited to Mystere-Falcon 20-C5 aircraft Serial Number 141. See: Airplane Flight Manual Supplement 236A-F20-34.80.02AFMS Revision N/C, Maintenance Manual Supplement 236A-F20-34.80.02MMS Revision N/CO later approved revisions. | 01.04.2008 |
| 10015125 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03475 | REV. 1 | | FALCON 20 | Dual Collins TDR-94D Enhanced Mode S Transponder Upgrade. | | Master Document List 236B-MDL Revision Original, Engineering Order 236B-F20-34.80.05EO Revision Original, Airplane Flight Manual Supplement 236B-F20-34.80.05AFMS Revision Original | This STC is limited to Mystere-Falcon 20-C5 aircraft Serial Number 141. See: Airplane Flight Manual Supplement 236B-F20-34.80.05AFMS Revision Original or later approved revisions. | 31.03.2008 |
| 10015259 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03713 | REV. 1 | | DC-9-87 (MD-87) | Cabin Layout Change/Modification of existing Iberia 109 passenger economy seat layout to 20 B/C + 79 E/C seat layout for S/N 49832; 49834; 49837; 49838; 49840; 49841 and 12 B/C + 89 E/C seat layout for S/N 49833; 49839; 49842 | | 1. Engineering Order 211D-EO Revision N/C from 10. Oct. 2008.2. Compliance Statements 211D-CS. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 02.12.2008 |
| 10015306 | RUAG AEROSPACE SERVICES GmbH | EASA.A.S.03897 | | | EMBRAER EMB-135BJ | Stone Flooring / Replacement of existing floor covering by a stone floor covering in the forward entrance area and in the rear lavatory. This modification will not affect the structural flooring, the stone floor covering is only a decorative component. | | Master Document List 289B-MDL Rev. A dtd. 25.May 2009 | EASA EWIS Generic Special Condition not applicable for this STC | 26.05.2009 |
| 10027158 | RUAG AEROSPACE SERVICES GmbH | P-EASA.A.S.03970 | | | CESSNA 525 | P-RNAV Upgrade for the Honeywell GNS-XLS Flight Management System | FAR-23 | -Engineering Order (including Continued Airworthiness Requirements): 295A-EO, Rev. A-Flight Manual Supplement: 295A-AFMS, Rev. N/C | Restricted to aircraft with S/N 0001 to 0359 | 10.09.2009 |
| 10027192 | RUAG AEROSPACE SERVICES GmbH | P-EASA.A.S.03963 | | | BD-700-1A10, -1A11 | EGPWS MK V Enhancement with RAAS Upgrade of the existing Honeywell Enhanced Ground Proximity Warning System (EGPWS) MK V with the Honeywell Runway Awareness and Advisory System (RAAS). Installation of an optional RAAS Advisory Mute switch into the center pedestal. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Document List, 291C-MDL Revision A dated 10 September 2009. Approval for Aircraft Modification, 291C-CN Revision A dated 10 September 2009 or later revisions of the above listed documents approved by EASA | BD-700-1A10: S/N 9002 and subsequent. BD-700-1A11: S/N 9127 and subsequent | 14.09.2009 |
| 10027824 | RUAG AEROSPACE SERVICES GmbH | | | | CL-600-2B16 | Installation of Honeywell Aircraft Flight Information System (AFIS) | EASA IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | RUAG Master Document List No. 078L-MDL Revision N/C dated 23/10/2009. RUAG AFMS No. 078L-AFMS Revision N/C dated 20/10/2009. RUAG SoC ref. 078L-CN Revision N/C dated 23/10/2009 or later revisions of the above listed documents approved by EASA | None. | 09.11.2009 |
| 10028151 | RUAG AEROSPACE SERVICES GmbH | | REV. 1 | | EMB-135BJ | AXE4-869/2009-mk (297A-RFC): Electronic Flight Bag | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Cert Basis per TCDS IM.A.032 plus -CRI H-1/1324 "EWIS"-CRI F-28/1324 "EFB" The certificate (...) | Master Document List 297A-MDL Rev. B dtd. 7.5.2010 - AFM Supplement 297A-AFMS Rev. N/C dtd. 23.10.09 - AMM Supplement 297A-AMMS Rev. A dtd. 7.5.2010 or later revisions of the above listed documents approved by EASA | None. | 30.06.2010 |
| 10030045 | RUAG AEROSPACE SERVICES GmbH | | | | CL-600-1A11 / -2A12 / -2B16 | EASA AD 2009-0097 and AD 2009-0173 for Transponder TDR-94D | EASA IM.A.023. The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | RUAG SoC reference 316A-CN Revision N/C dated 07 May 2010. RUAG MDL No. 316A-MDL Revision N/C dated 07 May 2010 or later revisions of the above listed documents approved by EASA | None. | 17.05.2010 |
| 10030478 | RUAG AEROSPACE SERVICES GmbH | | | | BD-700-1A11 | Stone Flooring | EASA IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | RUAG Master Document List No. 313B-MDL Rev. N/C dated 14 June 2010. RUAG Approval for Aircraft Modification No. 313B-CN Rev. N/C dated 14 June 2010. RUAG Airplane Maintenance Manual Supplement No. 313B-AMMS or later revisions of the above listed documents approved by EASA | None | 17.06.2010 |
| 10031258 | RUAG AEROSPACE SERVICES GmbH | | | | DC-9-83 (MD 83) | Cabin Layout/Interior Change -DC-9-83 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-001/SC H-001 | RUAG Doc. No. 293B-SD; 293B-MDL; 293B-AMMS or later revisions of the above listed documents approved by EASA | none | 05.08.2010 |
| 10032060 | RUAG AEROSPACE SERVICES GmbH | | | | CL-600-1A11 CL600-2B16, CL-600-2A12 | Installation of Satcom Iridium System Simphone DUO | EASA IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged (...) | RUAG Master Document List No. 078R-MDL Revision N/C dated 04 October 2010. RUAG AFMS No. 078R-AFMS Revision N/C dated 17 August 2010. RUAG SoC ref. 078R-CN Revision N/C dated 30 September 2010 or later revisions of the above listed documents approved by EASA | none | 06.10.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|------------------|----------------|----------------------|---|---|--|--|--|---------------|
| 10032061 | RUAG AEROSPACE SERVICES GmbH | | | | CL-600-1A11 CL600-2B16, CL-600-2A12 | Electronic Flight Bag NavAero tBagC2 (EFB) Class II | EASA IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | RUAG Master Document List No. 078S-MDL Revision N/C dated 16 September 2010. RUAG AFMS No. 078S-AFMS Revision N/C dated 01 October 2010. RUAG SCo ref. 078S-CN Revision N/C dated 04 October 2010. or later revisions of the above listed documents approved by EASA | none | 06.10.2010 |
| 10033641 | RUAG AEROSPACE SERVICES GmbH | P-EASA.A.S.03896 | | | EMB-135BJ | Modification 289A: Thrane & Thrane Aero SB Lite Satcom Installation | The Certification Basis for the original product remains applicable to this certificate/ approval. Cert Basis per TCDS IM.A.032 plus: CRI F-29/03896 "WLAN" Voluntary compliance to CRIH-1 "EWIS" The certificated noise and/ or emissions levels of the origi (...) | Modification 289A "Thrane & Thrane Aero SB Lite Satcom Installation" Master Document List 289A-MDL Rev. D dtd. 26.01.11-AFM Supplement: 289A-AFMS Rev. N/C dtd. 16.10.09-AMM Supplement with Airworthiness Limitations 289A-AMMS Rev. C dtd. 24.01.11 or lat (...) | None | 01.02.2011 |
| 10034074 | RUAG AEROSPACE SERVICES GmbH | | | | 525 525A | Installation of High Speed SATCOM system with on-board WLAN Interface. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI F-101 (MOC) "WLAN Implementation in Cessna 525, 525A" The requirements for environment (...) | Engineering Work Order: RUAG Doc.No. 355A-EO, Rev.: - Flight Manual Supplement: RUAG Doc.No. 355A-AFMS, Rev.: - Maintenance Manual Supplement: RUAG Doc.No. 355A-AMMS, Rev.: - or later revisions of the above listed documents approved by EASA. | STC limited to S/N 525-0516 The WLAN clients (Handsets, Laptops, etc) are not covered by this STC. The usage of these components is requiring operational approval. | 04.03.2011 |
| 10036250 | RUAG AEROSPACE SERVICES GmbH | | | | BD 100-1A10 (CHALLENGER 300) | Installation of EGWS MK-V with Runway Awareness and Advisory System (RAAS). | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate. CRI-RAAS issue 3 dated 19/08/2011. The requirements for environmental protection and the associa (...) | Master Document List (MDL) 326A-MDL dated 22/08/2011- Airplane Flight Manual Supplement 326A-FMS dated 5/08/2011 or later revisions of the above listed documents approved by EASA. | None | 25.08.2011 |
| 10036633 | RUAG AEROSPACE SERVICES GmbH | | REV. 1 | | BD-700-1A10 BD-700-1A11 | Dual Electronic Flight Bag (EFB) Installation | EASA IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunch (...) | RUAG Master Document List No. 341A-MDL Rev. N/C dated 15 September 2011. RUAG Approval for Aircraft Modification No. 341A-CN Rev. N/C dated 15 September 2011. RUAG Airplane Maintenance Manual Supplement No. 341A-AMMS or later revisions of the above listed documents approved by EASA | None | 27.09.2011 |
| 10039078 | RUAG AEROSPACE SERVICES GmbH | | | | BELL 205A-1 | Title: Load Attachment Device BRD01 on Bell 205A-1 Description: The BRD01 is a cabin mounted device for attaching loads on Bell helicopters. It is used with opened sliding doors and can be mounted left- and righthand simultaneously. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-29, Amendment 2 (voluntarily chosen) The requirements for environmental protection a (...) | EASA approved Rotorcraft Flight Manual Supplement for Load Attachment Device BRD01 when installed in Bell Helicopters Model 205A-1 Series. Doc. No.: 349A-RFMS, Revision N/C Chapter 4- Airworthiness Limitations - of Maintenance Manual Supplement for Load (...) | None | 04.04.2012 |
| 10042290 | RUAG AEROSPACE SERVICES GmbH | | REV. 1 | | CL-600-1A11, CL-600-2A12 CL-600-2B16 | Flight Management System NZ-2000 - Upgrade to SW 6.1 | As per TCDS The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunchange (...) | RUAG Master Document List 133F-MDL Rev N/C dated 19 November 2012. RUAG Approval for Aircraft Modification 133F-CN Rev N/C dated 19 November 2012. RUAG Airplane Flight Manual Supplement 133F-AFMS Rev N/C dated 13 November 2012 or later revisions of the above listed documents approved by EASA | none | 21.11.2012 |
| 10042291 | RUAG AEROSPACE SERVICES GmbH | | | | CL-600-1A11, CL-600-2A12 CL-600-2B16 | ACSS TCAS II / 2000 Change 7.1 | As per TCDS The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product areunchange (...) | RUAG Master Document List 133G-MDL Rev N/C dated 19 November 2012. RUAG Approval for Aircraft Modification 133G-CN Rev N/C dated 19 November 2012. RUAG Airplane Flight Manual Supplement 133G-AFMS Rev N/C dated 13 November 2012 or later revisions of the above listed documents approved by EASA | none | 21.11.2012 |
| 10045833 | RUAG AEROSPACE SERVICES GmbH | | | | DC-9-81 (MD-81) DC-9-82 (MD-82) DC-9-83 (MD-83) | Cargo Seat Bag Installation | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - Special Condition(s): CRI D-01-(0010022580) The following paragraph(s) at a later amendment: CS 25.853 Amdt 12 The req (...) | Master Document List 411A-MDL Rev. N/C dtd. 18.07.2013- AFM Supplement 411A-AFMS Rev. N/C dtd. 14.06.2013 or later revisions of the above listed documents approved by EASA. | per AFM Supplement 411A-AFMS Rev. N/C dtd. 14.06.2013 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce (...) | 22.07.2013 |
| 10015012 | RUAG SCHWEIZ AG | EASA.A.S.03314 | | | BOMBARDIER DHC-6 SERIES 300 | Avionics system upgrade- In: Weather Radar System incl. ART-2000, AA-2012V, CM-2000; Garmin GMX-200 Multi Function Display system; Avidyne TAS-620 system incl. Heading Adapter AHR-500; Garmin GTX-330 Mode S Transponder system; Collins DME IND-42 (...) | | Master Document List No. 131435-00 of 30.01.2009 or later revision- Instructions for Continuous Airworthiness No. 131435-60 of 19.03.2008 or later revision- AFM Supplement No. 131435-50 of 14.03.2008 or later revision | 1. Reg.: T-741 S/N: 4662 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the a (...) | 20.04.2009 |
| 10017236 | RUAG SCHWEIZ AG | EASA.R.S.01415 | | | EUROCOPTER FRANCE SA 365N | Enhanced navigation system/Removed camera adaptation on existing mount and certification of the operators rack | | Master Documents List, Doc. No. 308-05001 Rev. 2 or latest approved Revision- RFMS, Doc. No. 308-05500 Rev.1 or latest approved Revision | T-771 S/N: 6146 Valid for this S/N. | 04.08.2008 |
| 10026920 | RUAG SCHWEIZ AG | EASA.R.S.01550 | | | | | | | | 01.07.2009 |
| 10027200 | RUAG SCHWEIZ AG | P-EASA.R.S.01582 | | | MD900 | TV Broadcasting Equipment Installation (camera adaptation on existing mount and certification of the operators rack) | CS-27 Amdt. 2 dated 17.11.2008 | -Rotorcraft Flight Manual Supplement (RFMS) Doc. No. 1872438-50-Installation Instruction Doc. No. 1872438-61 | The Installation TV Broadcasting Equipment is limited to: S/N 900-00131 | 14.09.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|------------------|----------------|----------------------|------------------|---|---|--|---|---------------|
| 10030985 | RUAG SCHWEIZ AG | | | | P 180 AVANTI II | In-Flight Entert. System Installation by installing a ROSEN/RosenViewVX™ moving Map and DVD System, an IR-Eye, One FLIGHT DISPLAY SYSTEM 10.2 inch LCD and two 7 inch LCD's and a FLIGHT DISPLAY SYSTEM Wireless audio transmitter for audio distribution to SENNHEISER HDR140 wireless Headsets. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List: Nr. 163206-00, Rev. 000, 16.07.10-Instruction for continued airworthiness: Nr. 163206-60, Rev. 000.02.10.09-Declaration of Compliance: Nr. 163415-4100, Rev. 000-Aircraft Flight Manual Supplement: Nr. 163206-50, Rev. 000, 21.06. (...) | None | 20.07.2010 |
| 10031322 | RUAG SCHWEIZ AG | EASA.R.S.01547 | | | 500 N | IRIS Camera Installation. Installed: L3 Avionics System Type A100 IRIS Camera System. | | #, Master Document List: RUAG Doc. Nr. 1864441-00#, Instructions for Continued Airworthiness: RUAG Doc. Nr. 1864441-60#, Rotorcraft Flight Manual Supplement: RUAG Doc. Nr. 1864441-50 | Serial number: LN088. | 02.06.2009 |
| 10034921 | RUAG SCHWEIZ AG | | | | PC-6 ALL SERIES | Avionics upgrade with Aspen EFD1000 Removed- Various Equipment (if installed) i.a.w. DIC 1898256-101 Installed- Aspen EFD1000 Electronic Flight Instrument System- Dual Garmin GNS430W COM/NAV/GPS- Garmin GMA 340 Audio Panel | FAR 23 Amdt:54 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. SCs or ESFs: For this change, voluntary compliance to FAR 23 Amdt 54 an (...) | Engineering Order: 1898256-01, dated 11.04.2011 Airplane Flight Manual Supplement: 1898256-50 dated 11.04.2011 Instructions for continued airworthiness: 1898256-60, dated 01.04.2011 or later revisions of the above listed documents approved by EASA. | S/N: All- One independent Communication or Navigation System needs to be installed in addition to the two GNS430W - RNAV(GPS) Approaches to LNAV, LNAV/VNAV or LPV minima are not approved. | 13.05.2011 |
| 10037011 | RUAG SCHWEIZ AG | | | | EC 130 B4 | Avionics Retrofit (Out: Garmin GTX-330 Mode S Transponder & GMA-340 Audio/Marker system; Honeywell KX 165A VHF-COM/NAV system; Aspen EFD 1000H EFIS system; Garmin GNS 430W VHF COM/NAV & GPS system; Garmin GSR 56 Iridium system; PS Engineering PMA 8000 (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. SCs or ESFs: Voluntary compliance CS 27 Amdt 0 for the change CS 27.1., 21.27.29.251 (...) | Flight Manual Supplement: FMS-1900810-01 Rev. Air later revisions of the above listed documents approved by EASA. Master Document List: MDL-1900810-01 Rev. A Instructions for Continued Airworthiness: ICA-1900810-01 Rev. A | S/N 7214 | 22.11.2011 |
| 10045837 | RUAG SCHWEIZ AG | | | | A109E | RUAG UNS-1Lw A109 Universal UNS-1Lw FMS and Thommen Air Data System AD32 Installation | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Rotorcraft Flight Manual Supplement for Agusta A109E with Universal Flight Management System (FMS) UNS-1Lw and Thommen AD32 Air Data Display, Reference 1894714-50, Revision 002, dated 15.07.2013 or later revisions of the above listed documents approved (...) | S/N 11106 | 23.07.2013 |
| 10014929 | RUECKER AEROSPACE GmbH | EASA.A.S.03182 | | | AIRBUS A340-300 | A340-300 new First Class Seats Test Aircraft / Installation of four newly developed First Class Seats in an Airbus A340-300 aircraft for testing. | | Master Document List, ref. Doc. No. MDL252-00001-000G, Issue G, dated 25.03.2008, Compliance Program, ref. Doc. No. CP_252-00001-000I, Issue I, dated 18.03.2008, Applicant's Declaration of Compliance - dated 25.03.2008 | | 25.03.2008 |
| 10014961 | RUECKER AEROSPACE GmbH | EASA.A.S.03232 | | | DORNIER 328-100 | Installation of a 230VAC Euro standard power socket at both cockpit sidewalls. | | MASTER Document List MDL-244-00200-000 Compliance document F_DO-019/B of 21 March 2007 Change Description numbered 6221070609 | | 22.08.2007 |
| 10027449 | RUECKER AEROSPACE GmbH | | | | DORNIER 228-101 | Installation of UHF Research Mission Equipment | The following new Certification Basis is applicable to this change CS-23, Amdt. 1, is elected to comply with for the items affected by the change. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | EASA approved Master Document List, Doc. No. MDL-890-00002-000, Issue D, or later approved issue | Limited to serial number 7083 | 01.10.2009 |
| 10027451 | RUECKER AEROSPACE GmbH | | | | 208B | Installation of UHF Research Mission Equipment | The following new Certification Basis is applicable to this change CS-23, Amdt. 1, is elected to comply with for the items affected by the change. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | EASA approved Master Document List, Doc. No. MDL-890-00001-000, Issue E, or later approved issue. EASA approved Aircraft Flight Manual Supplement, Doc. No. AFMS-890-00001-000A | Limited to serial number 708 | 01.10.2009 |
| 10027825 | RUECKER AEROSPACE GmbH | P-EASA.A.S.03855 | | | B777-200 | Installation of 2 single B/E Aerospace DS9000 steam ovens and 2 OCM9000 Oven Control Modules per B/E Aerospace Engineering Order EO-253-00004-001 Rev. D, or later approved revision. | The Certification Basis for the original product remains applicable to this certificate/ approval. CRI H-01 Enhanced Airworthiness Programme for Aeroplane Systems - ICA or EWIS for oven installation | B/E Aerospace Aircraft Maintenance Manual Supplement 43208-SM-01-B/E Aerospace Component Maintenance Manual for basic part number 4329102 | None | 09.11.2009 |
| 10029687 | RUECKER AEROSPACE GmbH | P-EASA.A.S.03810 | | | B747-400 | Installation of 2 single B/E Aerospace DS9000 steam ovens and 2 OCM9000 Oven Control Modules per B/E Aerospace Engineering Order EO-253-00002-001 Rev. A, or later approved revision | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-01 Enhanced Airworthiness Programme for Aeroplane Systems # ICA or EWIS for oven installation. | B/E Aerospace Aircraft Maintenance Manual Supplement 43208-SM-01B/E Aerospace Component Maintenance Manual CMM 25-35-750r later revisions of the above listed documents approved by EASA | None. | 13.04.2010 |
| 10030032 | RUECKER AEROSPACE GmbH | | | | SEE ATTACHED AML | Fullsize/ Halfsize ATLAS trolley installation approval. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Engineering order EO-253-00007-001, Issue B Drawing List DRW-253-00007-001, Issue B Trolley Stowage Information DE-253-00007-001, Issue A | See Interface Control Drawing AIS-09-03, Issue C. | 13.05.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|----------------|----------------|----------------------|---|---|---|--|---|---------------|
| 10034669 | RUECKER AEROSPACE GmbH | | | | 328-300 | Modification to certificated LOPA | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Declaration of Compliance of Ruecker Aerospace Entwicklungsbetrieb EASA.21J.296 dated 18.04.2011 | none | 19.04.2011 |
| 10038569 | RUECKER AEROSPACE GmbH | | REV. 1 | | 777-200 | Ruecker Aerospace Nr.: 102100040: Installation of CAIR (TM) system | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval-CRI H-1 ICA on EWS | Master Document List MDL-217-00001-000A Iss. A dtd. 22.Feb 2012-ICA per AMM Supplement AMS-217-00001-000A Iss. A dtd. 16.Feb 2012 or later revisions of the above listed documents approved by EASA | This modification is limited to MSN 29953. | 23.11.2012 |
| 10046549 | RUECKER AEROSPACE GmbH | | | | DC-9-83 (MD-83) | MD 83 stowage of items in FWD cabin | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Ruecker Aerospace Master Document List No. MDL-252-00043-000 Issue Adated 23 September 2013. or later revisions of the above listed documents approved by EASA. | Ruecker Aerospace Modification No. EO-259-00002-000 is a prerequisite for this STC. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change (...) | 30.09.2013 |
| 10017195 | S.A.M. - S.R.L. - | EASA.R.S.01333 | | | EUROCOPTER FRANCE AS 355F1 EUROCOPTER FRANCE AS 355F2 | SAM/002/2005: Archimede System "Digital acquisition and transmission of visible and infrared images" | | Flight Manual Supplement No. FMS/SAM-335/002_355 | | 09.11.2006 |
| 10017196 | S.A.M. - S.R.L. - | EASA.R.S.01334 | | | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 BA | SAM/002/2005: Archimede System "Digital acquisition and transmission of visible and infrared images" | | Flight Manual Supplement No. FMS/SAM/002_350 | | 09.11.2006 |
| 10014304 | S.E.A.M. | EASA.A.S.01548 | | | BEECH S35 | Installation of following equipments: KN63+KD1572+KX165+KI203+KMD150+AR4401+WX500+ANT NY163 | | FAM No 299/03 and AFMS for KMD 150 No 2 - Revision 0 | This STC is applicable only for the Beechcraft S35 - S/N D7498 | 16.09.2005 |
| 10014650 | S.E.A.M. | EASA.A.S.02805 | | | MOONEY M20M | BRNAV capacity modification - Installation of annunciator MD 41, technical description in FAM No. 1147/06/SEAM | | Airplane Flight Manual Supplement No. 1/SEAM dated 23rd May 2006. | Modification limited to s/n Z7-0135 | 16.03.2007 |
| 10014745 | S.E.A.M. | EASA.A.S.02928 | | | BEECH V35B | Modification d'installation BRNAV. | | FAM 1348/06/SEAM AFM Supplement n° 1, revision 0, dated 8 December 2006 | AFM Supplement n°1, revision 0, dated 8 December 2006 or any later approved revision must be on board. | 14.07.2008 |
| 10014787 | S.E.A.M. | EASA.A.S.02987 | | | PIPER PA-24-260 | Installation modification B-RNAV Mid Continent MD41-1464A. | | FAM SEAM No. 1144/06.AFM Supplement n° 1, revision 0, dated 22-05-2006. | Limited to S/N 24-4847 F-GKDA. | 07.11.2007 |
| 10014795 | S.E.A.M. | EASA.A.S.02997 | | | BEECH 100 | Installation of 2 GNS430A, 2 GTX330D XPDR, 1KMR 675 and 1 GAD42. | | Installation must be in accordance with S.E.A.M. MDL 1440-06/MDL/SEAM Ed.:1, Rév. 2, dated 29.06.07 or later Approved Revisions. Operation must be in accordance with S.E.A.M. Aircraft Flight Manual Supplement Supplément au Manuel de vol for GNS430 S (...) | Installation Limited to Beech B100, S/N. B-199 (F-GEXV). | 20.07.2007 |
| 10014796 | S.E.A.M. | EASA.A.S.02998 | | | KING AIR 200 | Installation of 2 GNS430A, 2 GTX330D XPDR and 1 GAD42. | | Installation must be in accordance with S.E.A.M. MDL 1428-06/MDL/SEAM Ed.:1, Rév. 2, dated 29.06.07 or later Approved Revisions. Operation must be in accordance with S.E.A.M. Aircraft Flight Manual Supplement Supplément au Manuel de vol for GNS430 S (...) | Installation Limited to Beech B200, S/N. BL-11 (F-GRSO). | 10.07.2007 |
| 10014797 | S.E.A.M. | EASA.A.S.02999 | | | BEECH 200 | Installation of 2 GNS430A, 2 GTX330D XPDR, 1KMR, 1KMR 675 and 1 GAD42. | | Installation must be in accordance with S.E.A.M. MDL 1441-06/MDL/SEAM. Ed. 1, Rev. 1, dated 04-07-2007 or later approved revisions. Operations must be in accordance with S.E.A.M. Aircraft Flight Manual Supplement for GNS430 SEAM AFM 1441-06-01, Rev. (...) | Installation limited to Beech B200, S/N BB-622 (F-GHSV). | 26.07.2007 |
| 10014842 | S.E.A.M. | EASA.A.S.03059 | | | BEECH B200 | Installation of one GTX330D and one TDR94. | | Installation must be in accordance with S.E.A.M. MDL 1587-07/MDL/SEAM Ed. 1, Rev. 0 dated 28.05.07 or later Approved Revisions. Operation must be in accordance with S.E.A.M. Aircraft Flight Manual Supplement for GTX330D et TDR94D Rev.0 dated 28.03.07 or later approved revision. | Limited to Beechcraft B200 SN BB-1116, F-GDJS. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effects (...) | 31.05.2007 |
| 10014845 | S.E.A.M. | EASA.A.S.03066 | | | F500 | Installation of a mode S GTX 330D Garmin Transponder. | | AFM Supplement SEAM-1588/07 dated 18 July 2007. | Limitation to s/n 500-0319. | 10.04.2008 |
| 10014939 | S.E.A.M. | EASA.A.S.03196 | | | PIPER PA-23-250 | Modification d'installation Radio/Couplage B-RNAV. Dépose: 2KX 175B+1 B.A. KMA 20+1 ADF KR85 + 1DLE KN65 + 1GPS 155XL + 1ATC KT 76 A. Pose: 1GNS 430 + 1KX 155 + 1B.Q. GMA 340 +1 ADFKR 87 + 1 DME KN 62A + 1 ATC GTX 330 + 1 MD41. | | Fiche d'approbation de conception d'une Evolution SEAM N° 1665-07/AFM supplément N°1 au manuel de vol, référence SEAM/SMV/1665/07 révision 0 dated 03/07/07. | Limited to S/N 27-4727 F-GOPR. | 07.11.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------|------------------|----------------|----------------------|-------------------|---|---|---|--|---------------|
| 10014944 | S.E.A.M. | EASA.A.S.03202 | REV. 1 | | PIPER PA-28R-201T | Installation modification B-RNAV. Installation of announcement unit Mid Continent MD 41-1484. | | 1581-07, Rev. 1 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 07.10.2008 |
| 10014970 | S.E.A.M. | EASA.A.S.03250 | | | BEECH F90 | Transponder ATC GTX330D, Radio/GPS GNS 530 and Radio/GPS GNS 430 installation. | | File modification No 1609-07/FM/SEAM. | Modification limited to SN LA-19.AFMS 1609-07-01, 1609-07-02 and 1609-07-03 must be on board. | 21.09.2007 |
| 10015070 | S.E.A.M. | EASA.A.S.03396 | | | PIPER PA-28R-200 | Modification d'installation Radio / B-RNAV sans couplage PADépote: 1 boîte audio KA 134+1 GPS 150XL+1 VHF/VOR KX 155 + IND KI 208 + 1MKRKR21 + 1 ATC KT 76APose: 1 boîte audio/MKR GMA 340 + 1 VHF/VOR/GLD/GPS GNS 430 + IND GI | | Fiche d'Approbation de Conception d'une Evolution N°1942-07- Révision 4 du 03/02/2009SUPPLEMENT N°1 AU MANUEL DE VOL - SEAM/SMV/1842/07 - Révision 2 du 03/02/2009 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2.Th (...) | 02.03.2009 |
| 10015095 | S.E.A.M. | EASA.A.S.03432 | | | CESSNA 210E | STC 1862-08 - Avionic modification for BRNAV navigation and autopilot coupling. | | AFM Supplement SEAM/SMV/1862/08 Revision 2, dated 28 April 2008. | Limited to S/N 58625. | 29.04.2008 |
| 10015105 | S.E.A.M. | EASA.A.S.03448 | | | SOCATA TB 20 | Installation of a GNS 430 W for IFR B-RNAV "en route" operations, without coupling to automatic pilot. | | - Flight Manual Supplement no. SEAM/SMV/1863/08 revision 0 dated 15/02/08 - SEAM Modification file no. 1863/08 Edition 1 dated 16/01/08 | 1. This installation is limited to TB 20 s/n 791.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no a (...) | 17.04.2009 |
| 10015135 | S.E.A.M. | EASA.A.S.03488 | | | BEECH 58 | STC 1940-08- Avionic modification for BRNAV navigation and autopilot coupling. | | AFM Supplement SEAM/SMV/1940/08 revision 1 dated 20 February 2008 | Limitation to s/n TH 1401 | 06.05.2008 |
| 10015196 | S.E.A.M. | EASA.A.S.03582 | | | PIPER PA-44-180 | Modification d'installation B-RNAV : pose Annonceur DWA 1000 | | 1) Fiche de Modification SEAM 2080/08/FACE/SEAM Edition 1 Révision 3 du 22/08/08Annexe C1 # Plan d'implantation des équipements +Annexe C2 # Plan électrique Edition 1 Révision 1 du 28/07/08Annexe C3 # Supplément No1 au manuel de volAnn | L'installation de ce STC est limitée au PA44-180 S/N 4479953081 L'utilisation du GPS en approche et sur SID's et STAR est interdite! Pré-requis: FAM 2004/08 #Pose GNS 430# suivant la modification approuvée EASA.A.C.096471 Avant installatio (...) | 30.09.2008 |
| 10015214 | S.E.A.M. | EASA.A.S.03629 | | | BEECH 200 | Modification d'installation transpondeur Mode S: dépose 2 transpondeurs KXP 755 + Boite de commande + 2 antennes ATC / Pose 2 transpondeurs GTX 330D + 4 antennes ATC | | Modification effectuée selon le document S.E.A.M « Fiche demodification 2125/08 », édition 1, révision 0, or later EASA approved revision. Aircraft Flight Manual Supplement document SEAM/SMV/2125/08, revision 0, du 30/06/2008, or later EASA approved revision. | 1. This modification is valid only for aircraft Beech 200, S/N BB 406.2.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modifi (...) | 05.08.2008 |
| 10015243 | S.E.A.M. | EASA.A.S.03678 | | | CESSNA 182R | MODIFICATION D'INSTALLATION B-RNAV avec couplage au pilote automatique | | Supplément manuel de vol SEAM/SMV/2127/08 Rev. 0 | Limitation to Cessna 182R S/N 182-68316 | 30.03.2009 |
| 10029007 | S.E.A.M. | | | | BEECH 58P | STC 2589/09 - Radio B-RNAV avec couplage PA | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:CS-23The certificated noise and/ or emissions levels of the original product are unchanged (...) | Supplément au manuel de vol SEAM/SMV/2789/09 rev 1 du 24/02/2010 or later revisions of the above listed documents approved by EASA | Limited to: S/N TJ-74 | 25.02.2010 |
| 10032905 | S.E.A.M. | | | | PA-28RT, PA-201T, | Avionic modification: dual installation of Garmin GNS 430W (with GA 35antennas) for B-RNAV operation with coupling to autopilot, installation of GMA 340 audio Intercom, KN 62A DME receiver and WX 500 stormscope with NY163 antenna | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | SEAM Flight Manual Supplement No 3049/10, Revision 1, dated 22 October 2010SEAM Modification File No 3049/10, Revision 1, dated 22 October 2010 or later revisions of the above listed documents approved by EASA | The installation is limited to aircraft PA-28RT/PA-28T S/N 28R813143.The aircraft must be equipped with a third radionavigation or radio communication mean independent from the two GNS 430 units.The GNS 430W software version must be No 3.3. Installation of a further version must be approved. | 07.12.2010 |
| 10033951 | S.E.A.M. | | | | FR182 | STC 3005/10 rev 0 - Modification d'installation des équipements radio et BRNAV avec couplage au pilote automatique. | The Certification Basis for the original product remains applicable to this certificate/ approval.CS 23The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Supplément Manuel de Vol SEAM/SMV/3005/10 Revision 0 dated 22 April 2010 or later revisions of the above listed documents approved by EASA. | Limitation to Cessna FR182 equipped with ASPEN EFD 1000 and NAVOMATIC 300A. | 22.02.2011 |
| 10035923 | S.E.A.M. | | | | C90 | Installation of a passenger information display, a Sigma-Tek 5000B-36attitude indicator, a Garmin GMA 340 audio panel, a Collins ADF-60Navigation system, a Garmin GTX 330 transponder, a Garmin G600 primary and navigation system with coupling to auto (...) | The Certification Basis for the original product remains applicable to this certificate/ approval.The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate (...) | AFM supplement Garmin G600 ref. SEAM/SMV/3381/10-2, Revision 2 dated 20/07/2011AFM supplement GNS 430 W ref. SEAM/SMV/3381/10, Revision 2 dated 20/07/2011SEAM modification sheet n° 3381/10, Revision 2 dated 20/07/2011 or later revisions of the above listed documents approved by EASA | This STC is limited to airplane s/n LJ-972.RNP APCH-VNAV and LPV Approach operations are not authorised.A third radionavigation or radio communication system independent from GNS 430 W units must be installed.Mechanical instruments must be installed t (...) | 27.07.2011 |
| 10036296 | S.E.A.M. | P-EASA.A.S.03809 | | | F90 | STC 2465/08 Rev 5GNS530W TAWS - GNS530W - Radar GWX 68 - TAS KTA 870 Installation. | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Dossier 2465/08 Rev 5 dated 18 July 2011 Supplément manuel de vol n° 1 SEAM/SMV/2465/08 Rev 5 dated 18 July 2011 Supplément manuel de vol n° 2 SEAM/SMV/2465/08 Rev 3 dated 9 October 2009 or later revisions of the above listed documents approved by EASA. | Limitation to Beech F90 equipped with SP2200 autopilot and modified according SEAM STC 2589/09 Dual G600 Installation. | 29.08.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------|------------------|----------------|----------------------|--|---|---|--|---|---------------|
| 10036367 | S.E.A.M. | P-EASA.A.S.03923 | | | F90 | STC 2589/09 Rev 4Dual EFIS G600 installation | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Dossier 2589/09 Rev 4 dated 18 July 2011 Supplement manuel de vol SEAM/SMV/2589/09 Rev 4 dated 18 July 2011 or later revisions of the above listed documents approved by EASA. | Limitation to Beech F90 equipped with SPZ200 autopilot and modified according to SEAM STC 2465/08 avionics installation. | 02.09.2011 |
| 10037304 | S.E.A.M. | P-EASA.A.S.03890 | | | PA-44-180 | Avionics upgrade including Aspen Electronic Flight Display (EFD) 1000Pro system and Garmin GNS 430 W with coupling to S-TEC 55X automatic pilot | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 23 original issue, § 303, 305, 561, 601, 603, 607, 609, 867, 1301, 109, 1311, 1321, (...) | SEAM Flight Manual Supplement SMV/2569/09/1, revision 4, dated 08/12/2009 and SMV/2569/09/2, revision 4, dated 20/11/09, and SMV/2569/09/3, revision 4 dated 27/11/09 or later revisions of the above listed documents approved by EASA | 1) This avionics installation if limited to aircraft S/N 44-8095025 and it is approved for BRNAV, PRNAV and RNP Approach/NAV operations. 2) The only authorized mode for GPS coupling to the S-TEC 55X automatic pilot (AP) is the NAV GPSS mode of the AP (...) | 17.11.2011 |
| 10037804 | S.E.A.M. | | | | 1900, 1900C, 1900D, 200, 200C | Installation of Garmin 500 Series system with coupling to autopilot Rockwell-Collins FCS 80, according to SEAM modification file n. 3541/11. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 2323.301, 23.303, 23.305, 23.307, 23.337, 23.341, 23.625, 23.1301, 23.1307, 23.1309, 2 (...) | Flight Manual Supplement no. SEAM-3541-AFM, revision 4 dated 09/12/2011. STC Data List no. SEAM-3541-SDL, revision 6 dated 20/12/2011 | Installation of this design change must be performed by SEAM technical personnel. If the GNS 500 Series system is installed whereas another Garmin GNS system is already installed on the airplane, a third independent radio communication or radionavigation (...) | 22.12.2011 |
| 10038255 | S.E.A.M. | | | | F177RG | Installation of a Garmin GNS 400 series system, a Garmin GA35 or GA 56GPS antenna, a Mid Continent MD41 annunciator and a Garmin GTX 330 transponder. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 2323.301, 23.307, 23.867, 23.1301, 23.1309, 23.1321, 23.1322, 23.1351, 23.1357, 23.1 (...) | SEAM Flight Manual Supplement No. 3623/11, revision 0 dated 02/09/2011, SEAM Modification File No. 3623/11, revision 2 dated 13/01/2012, or later revisions of the above listed documents approved by EASA. | Installation of this design change must be performed by authorised SEAM technical personnel. The aircraft must be equipped with at least one radionavigation or radio communication mean not similar to the GNS 400 series unit. This installation is approved (...) | 01.02.2012 |
| 10038454 | S.E.A.M. | | | | BN-2 | Installation of Garmin GNS 400 series system, GA 35 GPS antenna, GMX 200 multifunction display, GMA 347 audio box, SL30 NAV/COM system, GTX 330 transponder, Mid Continent MD 41 annunciator, Collins ALT 50 radial altimeter and Indigo One Tracking system, according to SEAM STC file No. 3622/11. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 23.301, 23.303, 23.305, 23.307, 23.337, 23.341, 23.365, 23.561, 23.625, 23.1301, 23.130 (...) | Flight Manual Supplement "GNS 400 SERIES B-RNAV on BN2" ref. SEAM-AFM-GNS, Revision 2 dated 26/07/2011. STC Data List ref. SEAM-SDL-BN2, Revision 3 dated 27/07/2011, or later revisions of the above listed documents approved by EASA. | This design change must be installed by SEAM technical personnel. This installation is approved for use in BRNAV operations. Coupling of GNS system to autopilot is authorised only for CENTURY 2000 autopilot. Installation of software updates must be approved (...) | 17.02.2012 |
| 10038702 | S.E.A.M. | | | | DR 400/180 | STC 3725/11 rev. 0 dated 04/01/2012 - Amélioration équipements radionav B-RNAV | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Supplément Manuel de Vol SMV 3725-11 rev. 0 du 04/01/2012 or later revisions of the above listed documents approved by EASA | S/N 2105 | 09.03.2012 |
| 10039279 | S.E.A.M. | | | | DA40D | Installation of two Garmin GNS 430 W systems, two GA 35 GPS antennas, a GMA 340 audiopanel, a GTX 330 Mode S transponder, a King KN-62A DME and Cobham antennas CI 292-2 (VHF), CI 105 (DME), CI 102 (MKR) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-23 The certified noise and/or emissions levels of the original product are unchanged (...) | Flight Manual Supplement No SEAM-FMV3723/11 revision 0 dated 28/11/2011 SEAM Modification File No 3723/11 revision 1 dated 30/12/2011 or later revisions of the above listed documents approved by EASA | Installation of this design change is limited to airplane S/N DA-002 and must be performed by authorised SEAM technical personnel. This design change is approved for B-RNAV operation without coupling of the GNS 430 W systems to an autopilot. The airp (...) | 23.04.2012 |
| 10040057 | S.E.A.M. | | REV. 2 | | AP68TP-300 (SPARTACUS) AP68TP-600 (VIATOR) P.68 B (VICTOR) P.68 C, P.68 C-TC P.68 OBSERVER, P.68 OBSERVER2 P.68 R (VICTOR), P.68 (VICTOR) P.68 TC (OBSERVER) | Installation of Aspen EFD 1000 Pro EFIS and Mid Continent MD41 annunciator according to SEAM modification n° 2622/11. Revision 2 extending applicability to all models of P68 aircraft type through SEAM modification n° 4001/12. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Data list n° SEAM 2622 SDL revision 3 dated 25/05/2012 Modification approval sheet n° 4001/12 revision 0 dated 03/12/2012 or later revisions of the above listed documents approved by EASA | Initial installation of this design change must be performed by authorised SEAM technical personnel. This installation is approved for BRNAV operations with coupling to Garmin GNS 430 or GNS 430 W system and to Century III autopilot. The modified airplane (...) | 12.03.2013 |
| 10040161 | S.E.A.M. | | | | B90, C90, E90 | Installation of a Garmin GTS 800 series Traffic Advisory System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-23 The certified noise and/or emissions levels of the original product are unchanged (...) | AFM Supplement No SEAM 3747/11 Revision 2 dated 14/06/2012 STC Data List No SEAM 3747/11 SDL Revision 4 dated 14/06/2012 or later revisions of the above listed documents approved by EASA | Initial installation of this design change must be performed by authorised SEAM technical personnel. The modified aircraft must be equipped with TAS antennas installed according to SEAM modification No. 2465/11 approved under EASA.10014783-A Garmin G600 (...) | 14.06.2012 |
| 10041160 | S.E.A.M. | | | | 56TC, A56TC, 58, 58A, G58, 95, 95-55, 95-A55, 95-B55, 95-B55A, 95-B55B, 95-C55, 95-C55A, D55, D55A, E55, E55A, B95, B95A, D95A, E95 | Installation of Garmin GNS 430 W system, GA 35 GPS antenna, GMA 340 audiopanel and Mid Continent MD 41 annunciator according to SEAM modification n° 3823/12. The certified noise and/or emissions levels of the original product are (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-23 Amdt 2 The certified noise and/or emissions levels of the original product are (...) | SEAM Flight Manual Supplement n° SEAM-SMV/3823/12, revision 0 dated 02/05/2012 SEAM modification file n° 3823/12 revision 0 dated 02/05/2012 or later revisions of the above listed documents approved by EASA. | Installation of this design change must be performed by authorised SEAM technical personnel. This installation is approved for BRNAV operations without coupling to an autopilot. Used of GPS for SID/STAR and approach procedures is not authorised. Software (...) | 24.08.2012 |
| 10043419 | S.E.A.M. | | | | B90, E90, C90, C90A, C90GT, C90GTI. | STC 3807/12 revision 3 - Installation STC EASA 10037574 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS 23 Amdt 2 The requirements for environmental protection and the associated certificate (...) | Supplément manuel de vol SMV 3807-12 Rev3 du 22 Jan. 2013 or later revisions of the above listed documents approved by EASA. | The STC is approved for aircraft equipped with EFIS/EHSI and autopilot defined in GARMIN IM 190-01007-A3 and 190-01007-02 and STC 2807/12 annex C2. Prior to installation of this design change it must be determined that the interrelationship between (...) | 28.01.2013 |
| 10044429 | S.E.A.M. | | | | 35-33, 35-A33, 35-B33, 35-C33 35-C33A, E33, E33A, E33C, F33 F33A, F33C, G33, H35, J35, K35 M35, N35, P35, S35, V35, V35A V35B, 36, A36, A36TC, B36TC | STC 4022/12 Revision 1 - Installation GNS430W Privilege BRNAV avec couplage au pilote automatique et EDM700 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. Refer to the paragraphs listed on FAM at CS23 amdt. The requirements for environmental protection and the associated certificate (...) | Supplément au manuel de vol EDM700 Rev. 1 du 28 janvier 2013 Supplement au manuel de vol GNS 430W Rev. 1 du 28 janvier 2013 or later revisions of the above listed documents approved by EASA | The STC is approved for aircraft equipped with automatic pilot and equipment defined in GARMIN IM 190-00356-08 dated May 2011 and up and STC 4022/12 annex C2. Prior to installation of this design change it must be determined that the interrelationship between (...) | 11.04.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|----------------------|--|---|--|---|---|---------------|
| 10045316 | S.E.A.M. | | | | B200, B200C, B200CGT, B200GT, B300, B300C. | 3924/12 rev1 - Installation systeme KHF 950 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 23 AmR3 - 301, 305, 307, 561, 601, 603, 607, 609, 1301, 1309, 1321, 1 (...) | Annexe C8 3924-12 Rev1 du 19/10/2012 Supplement au manuel devol SEAM-SMV 3924/12-Rev 1 du 19/10/2012 or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 17.06.2013 |
| 10046326 | S.E.A.M. | | REV. 1 | | PA-44-180 | Certification for P-RNAV operations and RNP LPV approaches of an existing avionics configuration based on Garmin GNS 430W system and G500EFIS | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraphs at a later amendment: 23.1301, 23.1309, 23.1321, 23.1322, 23.1329, 23.1353, 23.1525, 23.1581. The (...) | SEAM Flight Manual supplement n°SMV/3388/11 revision 2 dated 01/07/2013 SEAM modification file n°33811/11 revision 2 dated 01/07/2013 or later revisions of the above listed documents approved by EASA. | This STC is limited to aircraft s/n 447995308. By this STC, the aircraft is approved for BRNAV, PRNAV and RNP approaches operations with coupling of the Garmin GNS 430 W and G500 systems to the autopilot Altimatec ITC. Prior to installation of this design (...) | 12.09.2013 |
| 10029549 | S.E.I. SERVIZI ELICOTTERISTICI ITAL | P-EASA.R.S.01549 | | | MD HELICOPTERS 369HS | Installation of Emergency Locator Transmitter (ELT), Artex C406-2HM, P/N369-25543-101 | The Certification Basis for the original product remains applicable to this certificate/ approval | SEI RFMS - 369HS-1-1 at Rev A or later EASA approved revision-Report - SEI Description and CCL - ELT C406-2 Rev B dated 24 March 2010 or later EASA approved revision-Report - SEI Notice of Design Change - NDC 369-005/1- or later revisions of the above listed documents approved by EASA | Not Applicable | 31.03.2010 |
| 10016296 | S.I.P. TECHNICAL SERVICES | EASA.IM.A.S.02593 | REV. 1 | FAA STC ST09367SC | DC-9-81 (MD-81) DC-9-82 DC-9-83 | Installation of dual Canadian Marconi CMA 900 FMS with GPS sensor (This is a revision of the STC ST09367SC already validated by ENAC on 2 December 1997 and amended on 5 May 1998 to implement the PRNAV capability.) | | Airplane Flight Manual Supplement No. 3209-FM-001. Rev. C dated 20 April 2007. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 20.12.2007 |
| 10016297 | S.I.P. TECHNICAL SERVICES | EASA.IM.A.S.02594 | | FAA STC ST09389SC | DC-9-82 DC-9-83 | Installation of Single Canadian Marconi CMA 900 FMS with GPS Sensor. | | 1. AFM Supplement No. 3218-FM-001 dated 13 December 1997/2. Master Drawing List No. 3218-DL-001-501 Rev. B. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 05.03.2008 |
| 10016331 | S.I.P. TECHNICAL SERVICES | EASA.IM.A.S.02640 | | FAA STC NO. ST9949SC | BOEING DC-9-81 (MD-81) BOEING DC-9-82 (MD-82) BOEING DC-9-83 (MD-83) | Installation of required sensor and interface for existing DFDR and FDAU to meet FAR 121.344. | | Master Data List 3273-DL-001-501 Rev. D dated December 5, 2002 | | 10.06.2008 |
| 10030551 | S4A SOLUTIONS FOR AVIATION, S.L. | | | | EC135 P2+ | Installation of an Iridium Satellite Telephone on board | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Drawing List Ref 13504-RPT04, Edition 2, dated 19/01/2010 Service Bulletin Ref: 13504-SB01, Edition 2, dated 01/02/2010 Rotorcraft Flight Manual Supplement Ref: 13504-HDK01, Edition 2, dated 01/02/2010 or later revisions of the above listed documents approved by EASA | Not Applicable | 23.06.2010 |
| 10042171 | S4A SOLUTIONS FOR AVIATION, S.L. | | | | C-212-200 | "Installation of Iridium satellite phone". A satellite phone is added to the airplane. This satellite phone is not located in the cockpit and its used only by an operator located at the rear of the passenger cabin, according to "Plan de Certification A1203-CPL01", Edición 6, Date: 12/11/2012. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 1. "Plan de Certification A1203-CPL01", Ed 6, Date: 12/11/2012. 2. "Suplemento al Manual de Vuelo A1203-HDK01", Ed 3, Date: 10/05/2012. 3. "Suplemento al Manual de Mantenimiento A1203-HDK02", Ed 2, Date: 07/05/2012. 4. "Suplemento al Manual de Cable (...) | None. | 12.11.2012 |
| 10039017 | SA BERINGER | | | | 101 101A 101AP 101B 101BC 101D 101P | BRGAP-STC-003(0) dated 30 Jan 2012 - Replacement of wheel and brakesystem. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS22 The certified noise and/or emissions levels of the original product are unchanged (...) | Installation manual MM-STC-003(0) dated 30 Jan 2012; Nomenclature pieces NP-STC-003(0) dated 30 Jan 2012; Time Limits and maintenance checks manual MC-STC-003(0) dated 30 Jan 2012; or later revisions of the above listed documents approved by EASA. | None. | 30.03.2012 |
| 10041414 | SA BERINGER | | REV. 1 | | ASW 20 BL, ASW 20 C, ASW 20 CL, ASW 20 F, ASW 20 FL, ASW 20 B, ASW 20, ASW 20 L | BRGAP-STC-005(0) - Replacement of wheel and brake system | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | MC-STC-005 revision 0 Time Limits / Maintenance check or later revisions of the above listed documents approved by EASA | none | 19.09.2012 |
| 10041955 | SA BERINGER | | REV. 1 | | DR400-100/-120/-120A/-120D DR400-125/-125V/-140/-140B DR400-160/ 160D DR400-180/-180R/-180S DR400-200R/-2+2/-500 DR400-NGU/-RP R1180 T, R3000/100 R2100, R2100A, R2112, R2120U R2160, R2160D, R2160I R3000/120, R3000/120 D R3000/140, R3000/160 R3000/160 S, R3000/180 | BRGAP-STC-006 revision 7 - Replacement of brake unit. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - the following paragraph(s) at a later amendment: Refer to CS paragraphs listed on document MM-STC-006(7) The requirements (...) | Dossier de Modification DM-STC-006 revision 7 dated 11 April 2013; Installation and Maintenance Manual MM-STC-006(0) dated 05 Oct 2012; Time Limits / Maintenance checks MC-STC-006(0) dated 05 Oct 2012; or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 15.04.2013 |
| 10014488 | SAAB AB | EASA.A.S.02550 | | | BEECH A100 | Installation of avionics in accordance with Saab Aerotech AB, TDM06-04 | | AFM-supplement: TDM06-04-3-AFMS1 (Garmin GNS-530), TDM06-04-3-AFMS2 (Garmin GNS-430), AT5813, 2005-03-30, Beech A100, s/n B-217 (Sandel ST3400) | This STC is limited to Beech A100, s/n B-217 | 19.04.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|------------------|----------------|----------------------|--|--|---|---|---|---------------|
| 10014508 | SAAB AB | EASA.A.S.02582 | | | LEARJET 35A | Mission Power | | SAAB Aerotech Modification TDM06-01-1-CP | -- | 20.06.2006 |
| 10014936 | SAAB AB | EASA.A.S.03192 | | | CESSNA 560 | Replacement of SSR Mode S Transponders for compliance with Enhanced Surveillance. | | SAAB AB TDM07-18, revision defined in TDM07-18-O-LEP, rev. A, 2007/11/01.AFM Supplement TDM07-18-3AFMS, ref. A, 2007-08-15. | | 04.12.2007 |
| 10014937 | SAAB AB | EASA.A.S.03193 | | | CESSNA 650 | Replacement of SSR Mode S Transponders for compliance with Enhanced Surveillance. | | Saab AB TDM07-19, revision defined in TDM07-19-0-LEP, Rev. A, 2007-11-01.AFM supplement TDM07-19-3-AFMS, rev. A, 2007-08-15. | | 04.12.2007 |
| 10015123 | SAAB AB | EASA.A.S.03472 | | | MITSUBISHI MU-2B | Garmin GTX 330/330D (SSR Mode S Transponder) upgrades for compliance with Enhanced Surveillance (EHS) requirements. | | Saab AB TDM08-03, with revisions defined in TDM08-03-0-LEP, Rev. A, 2008-02-22AFM Supplement TDM08-03-3AFMS, Rev. A, 2008-01-28 | | 27.03.2008 |
| 10015162 | SAAB AB | EASA.A.S.03532 | | | LEARJET 35A | Mode S Enhanced Surveillance (EHS) Upgrade. | | AFM Supplement as per TDM08-02-03-AFMS, Issue A (or later approved revisions). | EHS Non Capable. | 01.07.2008 |
| 10029935 | SAAB AB | | | | MU-2B | Installation Provisions/Video Signal in Mitsubishi MU-2B rear pressurebulkhead | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS-23 Amdt. 1, dated 12 February 2009The requirements for environmental protection and (...) | TDM10-03 as described in TDM10-03-LEP Revision A dated 2010-04-30 | Target Towing provisions according to Swedish CAA approval C6A/71 mustbe installed. | 04.05.2010 |
| 10031209 | SAAB AB | P-EASA.A.S.03618 | REV. 2 | | 382 | Installation of Saab Missile Protection System. The installationconsists of a forward sensor fairing on both sides of the nose and a rear sensor installation in the tail. The dispensers are installed onboth sides under the rear edge of the wing.Insid (...) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalCS-25 Am 6The requirements for environmental protection and the associatedcertificated (...) | Installation In accordance with Saab Aerotech Modification refTDM07-37-ID rev B, dated 2010-02-15 or later approved revisions.Operation in accordance with Flight Manual TDM07-37-4-AFMS rev D, dated2010-04-19 or later approved revisions. To be Maint (...) | Aircraft Serial 4954 Only | 21.12.2010 |
| 10033697 | SAAB AB | | | | AS 355 E, AS 355 F, AS 355 F1 AS 355 F2, AS 355 N, AS 355 NP | Installation of New TopEye Mission Box | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS27 change 2Therequirements for environmental protection and the associatedcertificat (...) | TDM08-30-1-CP revision E dated 09 August 2010 - Certification Plan;TDM08-30-3-RFMS Flight Manual Supplement;TDM08-30-3-ICA revision B dated 10 May 2010 - Instructions for ContinuedAirworthiness | none | 03.02.2011 |
| 10038972 | SAAB AB | | | | C-212-EE | TDM10-05 Rev. G titled "Cabin Layout", dated 2012-03-20 "New Cabin Layout for Maritime Surveillance System in CASA C212". Installation and Mounting of Maritime Surveillance System and change ofCabin Layout to accommodate a maximum of 5 crewmembers an (...) | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Instructions for Continued Airworthiness TDM10-05-4-ICA Rev. B, dated2012-03-16 or later revisions of the above listed documents approved by EASA | All equipment necessary for Maritime Surveillance Missions shall beinstalled or added with EASA DOA 21J.108 Minor Change Approvals | 27.03.2012 |
| 10015116 | SAAB AB, AERONAUTICS | EASA.A.S.03465 | REV. 1 | | LEARJET 35A | Installation of Honeywell MK VIII EGPWS and Mission InhibitSystem. | | TDM08-06 with revision as described in TDM08-06-0-LEP, Rev. A, 2008-08-22. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 25.08.2008 |
| 10040413 | SAAB AB, NYKÖPING | | | | MU-2B-20, MU-2B-26 | Installation of Alt Pressure Bulkhead and Cabin Pressure Skin Feed Thruconnections for Antenna / Electrical Connectors | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.CS-23 amendment 2, dated 28 September 2010The requirements for environmentalprotection (...) | TDM11-15 as described in TDM11-15-0-LEP Revision 0 dated 11th June 2012Instructions for Continued Airworthiness: TDM11-15-4-ICA Revision Adated 21 st June 2011 | Mitsubishi MU-2B, S/N: 283, 287, 289, 293, 299, 320, 337 and 345 | 02.07.2012 |
| 10014330 | SAAB NYGE AERO AB | EASA.A.S.01583 | | | LEARJET 35A | Installation of Target Mission Operators Console | | Saab Nyge Aero Technical Directive-Modification, TDM03-03 | | 17.11.2005 |
| 10028070 | SABENA TECHNICS BOD | P-EASA.A.S.03815 | REV. 6 | | A340-311, A340-312, A340-313 | IAGOS ModificationThis STC installs the Integration of routine Aircraft measurements intoa Global Observing System. Theinitial Release of the STC was approving the technical provisions installing the IAGOS system.This Revision 1 approved the activati (...) | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.The requirementsfor environmental protection and the associatedcertificatednoise and/ (...) | Modification Approval Sheet, ref.: S-00-2071, Issue 7 dated 14 June2013Aircraft Flight Manual Supplement, ref.: AFM-S-F4030-01 Rev 03, dated 05June 2013Instruction for Continued Airworthiness, ref.: ICA-F4030-00-01 issue B,dated 30 May 2011(or later (...) | The modification is applicable to the following aircraft Serial Numbers:MSN 304, 327, 335, 347, 352, 354, 355, 377, 379, 390, 433, 434, and 447.Prior to installation of this modification it must be determined thatthe interrelationship between this m (...) | 18.07.2013 |
| 10032216 | SABENA TECHNICS BOD | | | | F28 MK0100 | Modification wireless telephones installation. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Certification plan DO-N-2009-174-PDE Issue 2 dated 15 September 2010AFM Supplement FK100-25-LAIKA-01-FM Issue K dated 29 March 2010(or later revisions of the above listed documentsapproved by EASA | PER AFM Supplement FK100-25-LAIKA-01-FM Issue K dated 29 March 2010Applicable only to MSN 11368, 11403 and 11406 | 18.10.2010 |
| 10032391 | SABENA TECHNICS BOD | EASA.A.S.03671 | REV. 2 | | A330-223 | A330-223 "P710" VIP Cabin CompletionThis STC installs a VIPcabin configuration.Revision 1 was issued to update the Limitation Section.Revision 2 is issued to capture a design evolution (9000 VU change) andassociated documentation update. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approvalEASA TCDS A.004CRI A-4301-002 EASA Supplemental Type Certification Basis The requirement (...) | Modification Approval Sheet, Ref: S-00-2046, Issue 7, dated25 April.2012Airplane Flight Manual Supplement, Ref: AFM-S F3028 MOD00-2046, Rev D,dated 24 April 2012.Instructions forContinued Airworthiness, Ref: MPDS-P710, Issue 4 from02 Nov. 2010Damage (...) | None | 26.04.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------|----------------|----------------|----------------------|--|--|---|---|---|---------------|
| 10033032 | SABENA TECHNICS BOD | | | | A318 ALL MODELS A319 ALL MODELS A320 ALL MODELS A321 ALL MODELS | Full provision for EFB Class 2 on Single Aisle Aircraft | Certification Basis in accordance with EASA Type Certificate Data Sheets A.064. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:1. CRI H- (...) | - Modification Approval Sheet, ref. Modification S-46-2102. Issue 2, dated 10th December 2010- Certification Program, ref. DO/N/2009/226/DZ, Issue 3, dated 10th December 2010. Airplane Flight Manual Supplement, ref. AFM-S-D9032-01, Rev. Cor later revis (...) | Not applicable | 15.12.2010 |
| 10033281 | SABENA TECHNICS BOD | EASA.A.S.03943 | | | F28 MK 0100 | Fokker 100 CABIN VIP (premod EASA STC A.S.02677) | JAR 25 Change 11 | CRI D-04 dated 7 January 2011 AFM Supplement FMS-FK-1003-25-001 dated 7 January 2011 | This Change to STC A.S.02677 does not require a technical change to the aircraft Private use only, max 30 passengers | 11.01.2011 |
| 10037593 | SABENA TECHNICS BOD | | REV. 2 | | A340-542 | VIP Tunisair upgrading project This STC installs a new cabin configuration, with presidential area. The revision 1 updated the Associated Technical Documentation, in line with the design evolution. This revision 2 updates some Certification documents, following the activation of specific system. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: See CRIA-1301-002 The requirements for environmental protection and the associated certified noise and/or emissions level (...) | Modification Approval Sheet, Ref: MAS S-00-2156 Revision 6, dated 16 Sep 2013 Technical Repercussion Sheet, Ref: TRS S-00-2156 Revision D, dated 12 Mar 2012 Aircraft Flight Manual Supplement, Ref: AFMS-F4032-00-001 Revision G, 14 Mar 2012 Airworthine (...) | This STC is applicable to MSN 902 MMEL Supplement associated with this STC is not covered by this STC approval. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and any other previous (...) | 25.09.2013 |
| 10040867 | SABENA TECHNICS BOD | | REV. 2 | | A320-214 | Structural provisions for the installation of SATCOM Antenna AMT3800. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Certification Plan PC-D0088-001, Issue 8, dated 17/09/2013- Master Drawing List MDL-S-53-2258-1, Issue D, dated 20/07/2012- Master Drawing List MDL-S-53-2258-2, Issue B, dated 06/11/2012- Instructions for Continued Airworthiness ICA-D0088-00-002, (...) | This approval is limited to MSN 764 and MSN 3164. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noa (...) | 20.09.2013 |
| 10040868 | SABENA TECHNICS BOD | | | | A320-214 | Installation of SATCOM system | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Certification Plan PC-D0088-002 Issue 4 dated 20/07/2012- Master Drawing List MDL-S-00-2261 Issue B dated 26/07/2012- Instructions for Continued Airworthiness (ICA) ICA-D0088-00-001 Issue A dated 26/06/2012- Modification Approval Sheet MAS-S-00-2261 I (...) | This modification can be installed only concurrently with the modifications approved under EASA STC 10040867 (EASA project0010016700). This approval is limited only to MSN 3164. | 31.07.2012 |
| 10043211 | SABENA TECHNICS BOD | | | | A320-214 | Installation of new 9101VU & 9102VU racks in cargo area | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Certification Plan PC-D0087-003 Issue 3 dated 07/01/2013- Master Drawing List MDL-S-53-2269-01 Issue D dated 03/01/2013- Instructions for Continued Airworthiness (ICA) ICA-D0087-00-002 Issue A dated 07/12/2012- Modification Approval Sheet MAS-S-53-2269 Issue 4 dated 07/01/2013 or later revisions of | This approval is limited only to MSN 764 | 11.01.2013 |
| 10043212 | SABENA TECHNICS BOD | | REV. 1 | | A320-214 | Structural provisions for Live TV Antenna. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Certification Plan PC-D0087-001, Issue 7, dated 16/09/2013- Master Drawing List MDL-S-53-2270, Issue D, dated 03/01/2013- Instructions for Continued Airworthiness ICA-D0087-00-003, Issue B, dated 05/09/2013- Airworthiness Limitation Supplement ALS- (...) | This approval is limited to MSN 764. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 20.09.2013 |
| 10043218 | SABENA TECHNICS BOD | | | | A320-214 | Installation of new forward INMARSAT SATCOM system, aft SATCOM SBB upgrade, new Live TV system and upgrade of the IFE and CMS | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Certification Plan PC-D0087-002 Issue 3 dated 07/01/2013- Master Drawing List MDL-S-00-2230 Issue A dated 07/01/2013- Instructions for Continued Airworthiness (ICA) ICA-D0087-00-001 Issue A dated 06/12/2012- Modification Approval Sheet MAS-S-00-2230 (...) | This approval is limited only to MSN 764. This modification can be installed only concurrently with the modifications approved under EASA STC 10043212 (EASA project0010016696). | 11.01.2013 |
| 10043408 | SABENA TECHNICS BOD | | REV. 2 | | ATR 42-500 | Cabin and Communication systems upgrade. This modification consists in installing SATCOM and cabin telephony capacity, as well as a modification of the rear galley area (new galley and seat arrangement reconfiguration). The Revision 1 relaxed the initial (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Modification Approval Sheet, ref: MAS-S-00-2282 Issue 7, dated 22 April 2013 Aircraft Flight Manual Supplement, ref: AFMS-S4014-00-001 Issue 3, dated 17 January 2013 Instruction for Continued Airworthiness, ref: ICA-S4014-00-011 Issue A, dated 20 Decem (...) | The embodiment of this STC is limited to MSN 859. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noad (...) | 24.07.2013 |
| 10045336 | SABENA TECHNICS BOD | | REV. 1 | | 737-700 | SATCOM upgrading, IFE/CMS, WIFI and GSM new systems modification | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: - CRI F-01 on IFE, issue 2 dated 6 June 2013 - CRI F-02 on Digital Devices Des (...) | - Certification Program, Reference DO/N/066/2012/PHG issue 6, dated 12 June 2013 - CRI F-01 on IFE, issue 2 dated 6 June 2013 - Certification Matrix, Reference DO/N/066/2012/PHG dated 17 June 2012 - Airplane Flight Manual Supplement, Reference AFMS-B3024-00-050 issue 2, dated 14 Ju (...) | Limitations: 1) The GSM system shall not be used below 10 000 feet pressure altitude 2) The ACARS system using the SATCOM shall not be used during any flight phase nor on ground. 3) The deployable IFE screens must be stowed during take-off and landing 4 (...) | 18.06.2013 |
| 10045901 | SABENA TECHNICS BOD | | REV. 2 | | A318-112 | Installation of cabin systems upgrade: - Satellite communication system avionics and antenna. - Satellite Live TV system - In Flight Entertainment and Cabin Management System - New humidifier system | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): CRI D-5310-010 "Glass Displays / Monitors" Deviation(s): CRI D-5305-010 "Firm handheld" The requireme (...) | Modification Approval Sheet MAS-S-00-2294, Issue 3, dated 24/07/2013 Master Drawing List MDL-S-00-2294, Issue B, dated 25/07/2013 Instruction for Continued Airworthiness ICA-D8004-00-01, Issue A, dated 02/07/2013 Damage Tolerant Airworthiness Limitation (...) | The approval is limited to MSN 2910. The approval is limited to 2914 flight cycles due to temporary fatigue limit. | 30.07.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------------|----------------|----------------------|---|--|---|---|---|---------------|
| 10045974 | SABENA TECHNICS BOD | | | | A330-301/-302/-303/-321/-322 A330-323/-341/-342/-343 | IAGOS Modification This STC installs the Integration of routine Aircraft measurements into a Global Observing System on A330-300 Series. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Modification Approval Sheet, ref. : S-00-2275, Issue 2 dated 29 July 2013-Aircraft Flight Manual Supplement, ref : AFM-S-F3034-00-01 Issue 1, dated 05 July 2013-Instruction for Continued Airworthiness, ref : ICA-F3034-00-05 issue A, dated 16 May 2013or (...) | The modification is applicable to the following aircraft Serial Numbers: MSN 421 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and (...) | 01.08.2013 |
| 10046071 | SABENA TECHNICS BOD | | | | A319-133 | Installation of a Cabin System Upgrade-., In Flight Entertainment system-., Cabin refurbishing-., Cabin communication system | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Modification Approval Sheet MAS-S-00-2310, Issue 5, dated 09/08/2013; Master Drawing List MDL-S-00-2310, Issue B, dated 06/08/2013; Instructions for Continued Airworthiness ICA-DO46-00-001, Issue A, dated 02/07/2013; Airplane Flight Manual Supplement A (...) | The approval is valid for A319-133CJ MSN 1468 only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no (...) | 12.08.2013 |
| 10014573 | SABENA TECHNICS DNR | EASA.A.S.02677 | | | FOKKER F.28 MARK 0100 | Fokker 100 VIP Cabin | | AFM Supplement FK-25-LAIKA-01-FM REV H | Maximum approved passenger capacity: 19 Limited to serial numbers 11368, 11403, 11406. Refer to AFM Supplement. | 03.07.2007 |
| 10043001 | SAFTGLO Inc. | EASA.IM.A.S.00543 | | FAA STC ST02248AT | A330-200 SERIES A330-300 SERIES | Installation of Saftglo Inc photoluminescent floor proximity emergency escape path marking system. (Validation of FAA approved STC ST02248AT) | | - MDL L-68-2 Rev. A dated Nov 15, 2000 or later FAA approved revision.- Flight Manual supplement L68-10 Rev I.R. | 1- The approval of this modification is limited to the identified aircraft models. The limitations and conditions defined in the EASA datasheet are also applicable. A copy of the STC approval and the MDL must be enclosed in the files of each modified a (...) | 27.01.2005 |
| 10015436 | SAF-T-GLO INC. | EASA.IM.A.S.01172 | | FAA STC ST01932AT | DC-9-31, DC-9-32, DC-9-34 | Installation of a photo luminiscent floor proximity emergency escape path marking system. | | - Flight Manual Supplement AFMS DAS report No. G49-11 IR, dated 7 July 1999.- Dispatch Operating Procedures DAS report No. G49-13 Rev. C dated 7 July 1999 or later approved revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 13.02.2009 |
| 10042972 | SAF-T-GLO INC. | EASA.IM.A.S.01173 | | | DC-9-51 | Installation of a photo luminiscent floor proximity emergency escape path marking system. | | Flight Manual Supplement AFMS DAS Report No. H75-11 Rev. IR dated 26 July 1999. Dispatch Operating Procedures DAS Report No. H75-13 Rev. IR dated 4 October 1999 or later approved revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.07.2005 |
| 10016732 | SAGEM AVIONICS, LLC | EASA.IM.R.S.01056 | REV. 4 | FAA STC SR09263RC-D | EC120B | SFIM two-axis VFR Autopilot System (SAGEM PA85) (Note: - Original STC: FAA STC SR09263RC-D, amended 9 February, 2011- Existing EASA STC: 10016732 rev 3- Revision 4 incorporates 2 new configurations related to minor changes to the FAA STC. | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 1. Installation of a SFIM VFR autopilot system in accordance with: AEC Master Drawing List 120A82-8010 revision E, dated June 02, 2006- Instructions for Continued Airworthiness PA85-120MMS-001 revision B, dated June 06, 2002, or later accepted revisions (...) | Installation of FAA STC SR09450RC (EASA.IM.R.S.01527) is mandatory for configurations -01 (<- -> -02) SAGEM Installation Drawing 120-1002 Installation of FAA STC SR09450RC (EASA.IM.R.S.01527) is mandatory for configurations -03, -04 of SAGEM Installation Drawing 120-1002 | 14.03.2011 |
| 10016737 | SAGEM AVIONICS, LLC | EASA.IM.R.S.01061 | | SR09291RC-D | EUROCOPTER FRANCE EC 130 B4 | Original STC nbr: FAA STC SR09291RC-D, amended 06-Oct-2006 Installation of a two-axis VFR auto-pilot | | - Definition and Installation: Heritage Aviation Ltd., Master Drawing List 130A82-8014 revision H, dated June 25, 2006, approved by FAA or later FAA approved revisions- Inspection and Maintenance: Instructions for Continued Airworthiness (mainte (...) | | 08.12.2006 |
| 10027509 | SAGEM AVIONICS, LLC | P- EASA.IM.R.S.01527 | REV. 2 | FAA STC SR09450RC | EC120B | Original STC: FAA STC SR09450RC, amended 11-May-2009 SAGEM Integrated Cockpit Display System (ICDS) | The Certification Basis for the original product remains applicable to this certificate. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate. | 1. Installation of a SAGEM ICDS-8A Integrated Cockpit Display System (ICDS) coupled to SAGEM autopilot per FAA STC SR09263RC-D (EASA.IM.R.S.01056) in accordance with: SAGEM Master Drawing List 120-0002, revision D, dated April 01, 2009. This configuration (...) | For night VFR operations, the helicopter must be fitted with a backup instrument permitting attitude control, such as a standby attitude indicator or a slip/skid indicator | 24.11.2009 |
| 10029542 | SANDEL AVIONICS, Inc. | P- EASA.IM.A.S.02918 | | FAA STC ST01930LA | S550 | Installation of Sandel Avionics SA4550 Primary Attitude Display and Sandel Avionics SN4500 Primary Navigation Display. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List ST12361LA-T-01 Rev. M from 14 August 2008 Airplane Flight Manual Supplement ST12361LA-T-10 Rev. Original from 04 June 2007 for SA4550 Airplane Flight Manual Supplement ST12361LA-T-30 Rev. Original from 30 October 2006 for SN4500 Instr (...) | None. | 30.03.2010 |
| 10017096 | SAS MAT AVIATION | EASA.R.S.01017 | | | EUROCOPTER FRANCE EC 120 B | Upgraded Avionics configuration | | Definition: - installation of the following equipment: GARMIN GNS 480 GPS/Com/NAV/ with remote ATC GTX32; MFD MX20; TCAD9900 BX; AHRs APIRS F201 with its magnetometer; PFD/ND ICDS2000 system; standby horizon 4300-433; intercom PMA 7000B; CD audio sys (...) | 1. Limited to aircraft EC120B s/n 1049, F-GRAE only. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.08.2005 |
| 10014320 | SAS TECHNICAL SERVICES | EASA.A.S.01571 | REV 1 | | B737-800 | Change of the cabin layout to Swedish National Air Medevac (SNAM) SAS Project number 737-2520-01 and SAS Project number 737-2520-02 | | SAS Project number 737-2520-01- SAS Technical Order MTO-255155E- SAS Technical Order MTO-255491A- SAS Flight Manual Supplement for Boeing 737-800 Supplement No. SAS-737-SNAM, Project No. SAS-737-2520-01- SAS drawing 737-800 Air Medevac (...) | 1. SAS Project Number 737-2520-01/n 28318, 28324, 28325, 28326, 28327, 28328, 30195, 30196, 30197, 30469, 30470, 30471, 32276, 32277 and 322782. SAS Project Number 737-2520-02/n 28318, 28323, 28324, 28325, 28326, 28327, 28328, 30194, 30195, (...) | 29.09.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|-------------------------------|--|---|---|--|--|---------------|
| 10016404 | SAVANNAH AIR CENTER, L.L.C. | EASA.IM.A.S.02733 | | FAA STC ST03528AT | BOMBARDIER BD-700-1A11 | Installation, Complete Custom Interior, s/n 9229 Configuration. | | FAA statement of Compliance dated 14 March 2008.Savannah AirCenter Master Drawing List Number G5K9229-MDL-2500001, Revision "A", dated December 1, 2007/Instructions for continued Airworthiness, Bombardier Global 5000 Document number G5K9229-IFCS-001, Revision "IR", dated November 5, 2007 | Restricted to ASN 9229. | 01.04.2008 |
| 10016536 | SAVANNAH AIR CENTER, L.L.C. | EASA.IM.A.S.02913 | | FAA STC SA02683AT, CAA UK AAN | GULFSTREAM IV | Installation of L3 Communications Digital Flight Data Recorder With Teledyne Digital Flight Data Acquisition Unit. | | To be operated in accordance with Approved Flight Manual supplement ref GIV1132-AFMS-001, dated July 9, 2003, or later approved revision | Only approved for Aircraft manufactured before 18 August 2000 | 03.06.2008 |
| 10028136 | SAVANNAH AIR CENTER, L.L.C. | | | FAA STC ST03818AT | BOMBARDIER BD-700-1A10 | Complete Custom Aircraft Interior - ASN 9287. | EASA IM.A.009 and EASA MOC CRI F-01The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA SoC ref. ST03818AT dated 29 October 2009.FAA SoC ref. ST03818AT dated 30 November 2009.Savannah Air Center Master Drawing List, XRS9287-MDL-2500001, Rev A dated 14 October 2009 or later approved revision.Instructions for Continued Airworthiness, B (...) | Restricted to ASN 9287 for a total of 15 passengers. | 04.12.2009 |
| 10031798 | SAVANNAH AIR CENTER, L.L.C. | | | FAA STC ST12210AT-T | BD-700-1A10 | Installation, Complete Custom Interior, ASN 9347 Configuration. | EASA TCDS No. IM.A.009The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approval.The requirements for environmental protection and the associated (...) | FAA SoC ref. ST12210AT-T dated 13 September 2010.FAA STC No. ST03870AT dated 13 August 2010.Savannah Air Center MDL_XRS9347-MDL-2500001, Rev. A dated 06 August 2010.Instructions for Continued Airworthiness, Bombardier Global ExpressDocument Number X (...) | Restricted to ASN 93471.Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2.WLAN system is compatible only to IEEE 802.11(b). | 14.09.2010 |
| 10032322 | SAVANNAH AIR CENTER, L.L.C. | | REV. 1 | FAA STC ST03881AT | BD-700-1A10 | Installation, Complete Custom Interior, S/N 9345 Configuration | EASA TCDS No. IM.A.009The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approvalCRI F-01 - Installation of Wireless SystemThe requirements for env (...) | FAA SoC ref. ST12210AT-T dated 21 October 2010.FAA STC No. ST03881AT dated 26 October 2010.Savannah Air Center MDL_XRS9345-MDL-2500001, Rev. A dated 13 September 2010.Instructions for Continued Airworthiness, Bombardier Global ExpressDocument Number (...) | Restricted to ASN 9345 and 1. Refer to the Limitations sections of the required AFM Supplement and to the following FCOMS.G862081 Rev NC - FCOMS - WLAN System.G862092 Rev A - FCOMS - Mid-Cabin Door2. WLAN system is compatible only to IEEE 802.11(b) | 29.03.2011 |
| 10032416 | SAVANNAH AIR CENTER, L.L.C. | | | FAA STC ST03628AT | BD700-1A10 | Installation, Complete Custom Interior, S/N 9260 Configuration | EASA TCDS No. IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product (...) | FAA SoC ref. ST10844AT-T dated 02 November 2010.FAA STC No.ST03628AT dated 20 October 2008.Savannah Air Center MDL_XRS9260-MDL-2500001, Rev. A dated 03 October 2008.Instructions for Continued Airworthiness, Bombardier Global ExpressDocument Number X (...) | Restricted to ASN 9260 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2. WLAN system is compatible only to IEEE 802.11(b). | 04.11.2010 |
| 10014558 | SCAI TECH | EASA.A.S.02657 | REV. 3 | | DR 300/108, DR 300/120 DR 300/125, DR 300/140 DR 300/180 R, DR 315, DR 340 DR 360, DR 380, DR 400/100 DR 400/120 D, DR 400/125 DR 400/120, DR 400/120 A DR 400/140, DR 400/140 B DR 400/160, DR 400/160 D DR 400/180 S, DR 400/NGL DR 400/180 R | SCAI-DR315, DR300-108, 120, 125, 140, DR340, DR360-FM-A, SCAI-DR380, DR300-180R-FM-A, SCAI-DR412-FM-B, SCAI-DR416-414-FM-B, SCAI-DR418-FM-B, SCAI-DR221, DR221B-FM-A.Extension of EASA.A.S.02657 to DR221 aircraft. | CS-23 | Supplément au manuel de maintenance SCAI-DR315, DR300-108, 120, 125-<(-,-)>-140, DR340, DR360-MIM-A. Supplément au manuel de maintenance SCAI-DR380, DR300-180R-MIM-A. Supplément au manuel de maintenance SCAI-DR412-MIM-B. Supplément au manuel de maintenance SCAI (...) | None. | 23.10.2009 |
| 10034639 | SCAI TECH | EASA.A.S.02943 | | | F150L, F150M, FA150L, FA150M | SCAI-CE150-FM-A Echappement accordé avec silencieux double flux. | | -AFM supplement SCAI-CE150-SUPMDV-A dated 4 May 2009.-Maintenance Manual supplement SCAI-CE150-MIM-A dated 4 May 2009. | Limitation to aircraft equipped with engine TCM 0-200-A and propeller McCauley 1A102/OCM6948. | 02.06.2009 |
| 10013928 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.00239 | REV. 1 | | BEECH 400A | Installation of a EGPPWS Mk. VIII system in accordance with the Engineering Order Doc.: 2017016-EO-02 Issue 2 or later EASA approved revision. | | | 1. AFM Supplement Document No. 2017016-AM-02 Issue 2 or later EASA approved revisions.2. Maintenance Requirements and Instruction for Continuous Airworthiness Doc. 2017016-CA-02. | 15.02.2008 |
| 10013944 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.00384 | | | BAE 125 SERIES 800B | Installation of an Universal CVR-30B Cockpit Voice Recorder in accordance with the Engineering Order Doc: 2022468-EO-01, Issue 1 or later EASA-approved Revisions. AFM Supplement Document No 2022468-AM-01 Issue 1 or later EASA-approved Revisions. Maint (...) | | | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 07.12.2004 |
| 10013971 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.00766 | | | SAAB SF340A | Installation of an ADF 2 Bendix King KDF 806 System. | | SAD "Description of System and Installation" Ref.: 2020789-DE-02, Issue 3, dated 22.10.2004.SAD "SAAB SF 340A approved Aircraft Flight Manual Supplement for ADF 2 Bendix/King KDF 806" ref.: 2020789-AM-01, Issue 4, dated May 2005 | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 08.06.2005 |
| 10013994 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01028 | | | LEARJET 55 | Installation of a Rockwell Collins ACAS (TCASII) 4000 with dual Mode S Transponders | | -Engineering Order 2019360-EO-01, Issue 3-Certification Documents List, 24.1.2005-AFMS Issue 2, January 2005 | 1. Certification Basis for this STC is CS 25 plus JAA TGL No 8 and JAA TGL No 13 | 31.01.2005 |
| 10013995 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01029 | | | LEARJET 55 | Installation of a L3 Digital Flight Data Recorder | | -Engineering Order 2019360-EO-02, Issue 1-Certification Documents List, 30.12.2004-AFMS Issue 1, October 2004 | Certification Basis for this STC is CS 25 | 31.01.2005 |
| 10013997 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01032 | | | LEARJET 55 | Installation of a Honeywell EGPPWS mark VIII (TAWS) | | -Engineering Order 2019360-EO-03, Rev. 2-Certification Documents List, 24.01.2005-AFMS Issue 1, November 2004 | Certification Basis for this STC is CS 25 plus JAA TGL No 12 "TAWS" | 31.01.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|----------------|----------------|----------------------|--|---|---------------------|--|---|---------------|
| 10014015 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01058 | REV. 1 | | AMD/BA FALCON 10 | Installation of Rockwell Collins TCAS-4000 incl. Dual Mode S Transponder TDR-94D in accordance with the Engineering Order Doc 2022256-EO-01 Issue 2 or later EASA approved revisions | | AFM Supplement ACAS Document No 2022256-AM-01A issue 2 or later EASA approved revisionsAFM Supplement Mode S Document No 2022256-AM-01B Issue 2 or later EASA approved revisionsInstructions for Continued Airworthiness Doc 2022256-CA-01 | | 30.05.2005 |
| 10014016 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01059 | REV. 1 | | AMD/BA FALCON 10 | Installation of a Kannad 406 ELT in accordance with Engineering Order Doc 2022256-EO-03, Issue 1 or later EASA approved revisions | | AFM Supplement Document No 2022256-AM-03 Issue 2 or later EASA approved revision Instructions for Continued Airworthiness Doc 2022256-CA-03 | | 14.05.2005 |
| 10014017 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01060 | REV. 2 | | AMD/BA FALCON 10 | Installation of a Multi Function Radar Display (MFRD) and EGPWS MK VIII in accordance with the Engineering Order Doc 2022874-EO-01 Issue 2 or later EASA approved revisions | | AFM Supplement MFRD Document No 2022874-AM-01A Issue 2 or later EASA approved revisionsAFM Supplement EGPWS MK VIII Document No 2022874-AM-01B Issue 3 or later EASA approved revisionsInstructions for Continued Airworthiness Doc 2022874-CA-01 | | 09.08.2007 |
| 10014025 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01091 | | | BEECH 200 BEECH 200C BEECH B200 BEECH B200C | Installation of a Universal TAWS system and related components. | | - EASA approved AFM Supplement Doc. 2021128-AM-01, issue 5, or later EASA approved issue- Description of Change 2021128-DE, Issue 3 | | 04.01.2007 |
| 10014026 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01092 | | | BEECH 200 BEECH B200 | Installation of GPS TNL-2101 I/O APPR+ | | -AFM Supplement 2021128-AM-02 Issue 1 or later EASA approved revision-Continued Airworthiness Instructions 2021128-CA-02 | | 11.02.2005 |
| 10014027 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01093 | REV. 1 | | BEECH 1900 BEECH 200C BEECH 300 BEECH 300LW BEECH B200C BEECH B300 BEECH B300C | 1. Installation of Dual Mode S Transponder MST-67A System with ELS capabilities (Installation prior to 31-Jan-2007)2. Installation of Dual Mode S Transponder MST-67A System with ELS and EHS capabilities (Installation on or after 31-Jan-2007) | | 1. For installations prior to 31-Jan-2007:- AFM Supplement 2021128-AM-03, Issue 1 or later EASA approved revision,- Continued Airworthiness Instructions 2021128-CA-03.2. For installations on or after 31-Jan-2007:- Engineering Order, Doc (...) | For installations on or after 31-Jan-2007, this change is limited to the configuration of aircraft / equipment type as listed under item 4. in Engineering Order, Doc. No. 2021128-EO-03, Issue 5, or later approved issue: | 05.02.2007 |
| 10014028 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01094 | | | BEECH 200 BEECH B200 | Installation of Multifunction Display MFD-640 & RP 1008 | | -AFM Supplement 2021128-AM-04 Issue 2 or later EASA approved revision-Continued Airworthiness Instructions 2021128-CA-04 | | 11.02.2005 |
| 10014036 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01109 | | | CESSNA 525 | Installation of Honeywell KMH 980 Multi-Hazard Awareness System including TCAS I and EGPWS (Class B). | | EASA-approved Flight Manual Supplement 2019974-AM-01, issue2, or later EASA-approved issue. | | 11.04.2005 |
| 10014037 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01110 | | | CESSNA 525 | Installation of WX-500 Stormscope System | | EASA-approved Flight Manual Supplement 2019974-AM-02, issue2, or later EASA-approved revision. | | 11.04.2005 |
| 10014038 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01111 | | | CESSNA 525 | Installation of Bendix/King KLN 94 GPS System. | | EASA-approved Flight Manual Supplement 2019974-AM-03, issue2, or later EASA-approved issue. | | 11.04.2005 |
| 10014050 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01132 | | | LEARJET 35 LEARJET 35A | Installation of Honeywell CAS 67A ACAS/TCAS II with dual MST-67A Mode S Transponder | | -Certification Documents List 2020220-02, 19.04.2005-AFMS 2020220-AM-02A, Issue 2, or later EASA approved issues-AFMS 2020220-AM-02B, Issue 2, or later EASA approved issues | Certification Basis: applicable requirements of CS25 plus JAA TGLs No 8 and 13 1. | 31.05.2005 |
| 10014091 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01189 | REV. 1 | | CESSNA 500 CESSNA 550 CESSNA 560 CESSNA S550 | Installation of Kannad 406 AF ELT | | -Engineering Order 2022014 - EO - 02 - AFM Supplement, Doc. No 2022014 - AM - 02 - Continued Airworthiness Report Doc. No2022014 - CA - 02. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 07.04.2005 |
| 10014092 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01190 | | | CESSNA 650 | Installation of FreeFlight 2101 I/O APPROACH PLUS GPS Navigation System | | - Scandinavian Avionics Engineering Order No. 2019034-EO-03, Issue No. 2, dated August 3, 2005 - Approved Airplane Flight Manual Supplement for FreeFlight 2101 I/O APPROACH PLUS GPS Navigation System | | 25.08.2005 |
| 10014093 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01191 | | | CESSNA 650 | Installation of Honeywell CAS 67A TCAS II System, includingDual MST 67A Mode S Transponders | | - Scandinavian Avionics Engineering Order No. 2015055-EO-01, Issue 1, dated October 8, 2004- Approved Airplane Flight Manual Supplement for HONEYWELL MST 67A MODE - S Transponder with PSS50 Control, Doc. No. 2015055-AM-01B- Approved Airplan (...) | | 25.08.2005 |
| 10014095 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01193 | | | BAE ATP | Installation of Universal Terrain Awareness Warning System (TAWS) | | -Engineering Order Doc 2024352-EO-01 Issue 2 or later EASA approved revisions-AFM Supplement Document No 2024352-AM-01 Issue 2 or later EASA approved revisions - Continued Airworthiness Document No 2024352-CA-01 Issue 1 | | 26.07.2005 |
| 10014096 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01194 | | | | Installation of Universal UNS-1L Flight Management System | | -Engineering Order Doc 2024345-EO-02 Issue 2 or later EASA approved revision-AFM Supplement Document No 2024345-AM-02 Issue 2 or later EASA approved revision- Continued Airworthiness Document No 2024345-CA-02 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 26.07.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|----------------|----------------|----------------------|---|--|---------------------|--|---|---------------|
| 10014111 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01216 | | | BEECH 300 BEECH B300 | Installation of Honeywell CAS 67A ACAS/TCAS II with Dual MST-67A Mode S Transponder | | EAS approved Flight Manual Supplement CAS 67A ACAS II System Doc No 2023423-AM-01 Issue 2 or later EASA approved issue EASA approved Flight Manual Supplement MST67A Mode S Transponder with PS 550 Control Doc No 2023423-AM-02 Issue 2 or later EASA approved issue | | 30.05.2005 |
| 10014150 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01270 | | | BEECH 200 BEECH 300 BEECH B200 BEECH B300 | EGPWS Installation on Airbus A300/A310 | | Top Master Drawing List 00846-JAA-MDL Rev. D. AFM Supplement DC113-0846 Rev. B. Maintenance Planning Document Supplement DC114-0846-00 rev. A. | Refer to AFM Supplement Ref. Dc113-0846 Rev B and to Maintenance Planning Document Supplement DC114-0846-00 rev A. | 28.04.2005 |
| 10014151 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01271 | | | BEECH 200 BEECH 300 BEECH B200 BEECH B300 | Installation of Dual GTX 330D Mode S Transponder System. | | AFM Supplement Dual GTX 330D Mode S Transponder Doc. No. 2026288-AM-01 Issue 3 or later EASA approved revisions. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 29.06.2005 |
| 10014162 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01294 | | | BAE 125 -800 SERIES | Installation of a Rockwell Collins TCAS 4000 with dual TDR-94D Mode S Transponder | | -Engineering Order Doc No 2025716-EO-01 dated 09.03.2005 Issue 1 or later EASA approved revisions-Airplane Flight Manual Supplement/TCAS 4000 Doc No 2025716-AM-01A issue 2 or later EASA approved revisions-Airplane Flight Manual Supplement/Dual TDR | | 26.07.2005 |
| 10014199 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01349 | REV. 1 | | BEECH 200 BEECH 300 BEECH 300LW BEECH B200 BEECH B300 | Installation of TAWS ST3400 system.Revision of the AFM Supplement related to a Minor Change to EASA.A.S.01349.Installation of TAWS ST3400 system.Revision of the AFM Supplement related to a Minor Change to EASA.A.S.01349. | | AFM Supplement No. 2022641-AM-03, Issue 3 or later EASA approved revision. Continued Airworthiness Instructions 2022641-CA-03.EASA approved Aircraft Flight Manual Supplement Doc. No. 2022641-AM-03, Issue 4, dated January 2007 or later approved (..) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer.This STC is approved (..) | 26.02.2007 |
| 10014200 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01350 | | | CESSNA 500 | Installation of Honeywell KMH 880 Multi-Hazard Awareness System | | Airplane Flight Manual Supplement for rinstallation of Honeywell KMH 880 Multi-Hazard Awareness System Doc No 2027746-AM-01 Issue 1,dated May 2005 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 02.01.2006 |
| 10014226 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01389 | | | CESSNA 300 SERIES CESSNA 400 SERIES | Installation of GNS 430A/530A COMM/NAV/GPS Systems | | Airplane Flight Manual Supplement Doc.: 2027438-AM-02A 'Garmin GNS-430B/530 VHF Communications Transceiver/VOR/ILS Receiver/GPS Receiver', Issue 1 or later EASA approved revisionsAirplane Flight Manual Supplement Doc. No.: 2027438-AM-02B 'Ground Comm (...) | none | 20.09.2005 |
| 10014242 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01451 | | | CESSNA 500 CESSNA 550 CESSNA S550 | Installation of GPS400 in center panel with GAD 42 adapter for roll steering | | Approved Aircraft Flight Manual Supplement for Garmin GPS 400 GPS System Doc. No. 2027746-AM-02 Issue 2, dated August 2005 or later approved revision | | 24.01.2006 |
| 10014245 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01455 | | | CESSNA 300 SERIES CESSNA 400 SERIES | Installation of GTX 330 Mode S Transponder | | Airplane Flight Manual Supplement Doc.: 2027438-AM-04 'GTX Mode S-Transponder' or later EASA approved revision | | 20.09.2005 |
| 10014295 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01537 | | | BEECH 400A (HAWKER 400 XP) | Installation of a Rockwell Collins TCAS 4000 with dual TDR-94D Transponder | | Engineering Order Doc. No. 2027113-EO-01 Issue 1 or later EASA approved revisions.Airplane Flight Manual Supplement Rockwell Collins TCAS 4000 Doc. No. 2027113-AM-01A Issue 2 or later EASA approved Revisions.Airplane Flight Manual Supplement Rockwel (...) | | 15.11.2005 |
| 10014313 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01561 | | | BEECH 400A (HAWKER 400 XP) | Installation of a L3 Digital FDR and Teledyne FDAU | | 1) Engineering Order Doc. No.2029534-AM-01 Issue 2 or laterEASA approved Revisions.2) Airplane Flight Manual Supplement, Doc. No.2029534-AM-01 Issue 2 or later EASA approved Revisions.3) Instruction for Continuous Airworthiness Doc. No.2029534-CA-01. | | 22.11.2005 |
| 10014333 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01587 | | | LETECKE ZAVODY L 410 UVP-E20 | Installation of Honeywell CAS 67A TCAS II including dual MST 67A Mode S Transponder | | -AFM Supplement Doc 2026472-AM-01A, Issue 2 or later EASA approved revision-AFM Supplement Doc 2026472-AM-01B, Issue 2 or later EASA approved revision-AFM Supplement Doc 2026472-AM-01C, Issue 2 or later EASA approved revision-AFM Supplement Doc 2026 (...) | | 22.02.2006 |
| 10014343 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01603 | | | PIPER PA-34-200 PIPER PA-34-200T PIPER PA-34-220T | Installation of KMD 850 Multi Function Display | | AFM Supplement Piper Model PA-34-200, -200T, 220TDoc. 2029789-AM-01, Issue 1 | | 03.04.2006 |
| 10014360 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01625 | | | BEECH A36 | Replacement of existing KI-525A HSI system with new KI-825 EHSI system | | AFM Supplement Doc. 2029169-AM-01, issue 2, December 2005, or later EASA approved issue | | 20.02.2006 |
| 10014406 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.01701 | | | BEECH A36 | Title: GARMIN GNS 430/530 upgrade for GPS approachDescription: Upgrade of existing GNS 430/530 for GPS approach and installation of annunciator panel | | Aircraft Flight Manual Supplement, Doc. 2031235-AM-01, issue 1 or later EASA approved issue | | 13.02.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|----------------|----------------|----------------------|--|---|---------------------|--|---|---------------|
| 10014435 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02484 | REV. 1 | | CESSNA 500, CITATION CESSNA 550 CITATION II CESSNA 560 (CITATION V) CESSNA S550, (CITATION S/II) | Sandel ST3400 TAWS Installation left hand and right hand | | Engineering Order 2031680-EO-01 Issue 1 or later approved RevisionsAircraft Flight Manual Supplement 2031680-AM-01 A Issue 4 or later approved RevisionsAircraft Flight Manual Supplement 2031680-AM-01 B Issue 1 or later approved Revisions | Certifications Basis of the Change: CS 25, additionally JAA Leaflet No. 12 (TAWS) | 02.04.2006 |
| 10014455 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02509 | | | BOMBARDIER SD-3-60-SHERPA BOMBARDIER SD-3-SHERPA | Collins TCAS 4000 with dual TDR-94D Transponder Installation | | Engineering Order 2026479-EO-01 Issue 2, dated 19.05.2006 or later approved RevisionsAircraft Flight Manual Supplement 2026479-AM-01A Issue 1, dated January 2006 or later approvedRevisionsAircraft Flight Manual Supplement 2026479-AM-01B Issue 1, dated January 2006 or later approved Revisions | Certification Basis of the Change: CS-25, additionally JAA leaflet No. 8 Rev. 2 (ACASII), JAA Leaflet No. 13 Rev. 1 (Mode S ELS) | 01.08.2006 |
| 10014469 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02530 | | | CESSNA 500, CITATION CESSNA 550 CITATION II CESSNA 560 (CITATION V) CESSNA S550, (CITATION S/II) | Installation of a Garmin GNS 530A and a Honeywell Marker Beacon KR21, for EU basic RNAV operations | | System Description 2032044-DE Iss. 2, Drawing List 2032044-DL Iss. 2, MoC list 2032044_AFM Supplements 2032044-AM-A Iss. 3 and 2032044AM-B Iss. 2 | None | 13.06.2006 |
| 10014496 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02559 | | | DORNIER DO 228-100 DORNIER DO 228-101 DORNIER DO 228-200 DORNIER DO 228-201 DORNIER DO 228-212 | Sandel ST3400 TAWS installationInstallation of Class A TAWS/RMI Indicator in the right hand instrument panel with a remote Annunciator and Switching unit on the left hand instrument panel. Single 3" indicator is used to show compliance | | AFM Supplements No. 2032096-AM-A and AM-B, Issue 1 | -- | 22.06.2006 |
| 10014497 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02560 | | | DORNIER DO 228-100 DORNIER DO 228-101 DORNIER DO 228-200 DORNIER DO 228-201 DORNIER DO 228-212 | Installation of Honeywell CAS 67A ACAS II System incl. DualMode S XPDR | | EASA approved:AFM Supplementn Doc. 2032095-AM-A, Issue 2AFM Supplementn Doc. 2032095-AM-B, Issue 2AFM supplementn Doc. 2032095-AM-C, Issue 1AFM Supplement Doc. 2032095-AM-D, Issue 1 for later EASA approved issues | | 12.06.2006 |
| 10014539 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02626 | | | | Universal TAWS Installation incl. MFD-640 | | Engineering Order 20031553-EO-01 Issue 2, dated 20.04.2006 or later approved RevisionsAircraft Flight Manual Supplement2031553-AM-01A Issue 2, dated April 2006 or later approved RevisionsAircraft Flight Manual Supplement 2031553-AM-01B Issue 2, dated April 2006 or later approved Revisions | Certification Basis of the Change: CS 25 Amendment 1, additionally JAA Leaflet No. 12 (TAWS) | 27.06.2006 |
| 10014540 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02627 | | | | Installation of Universal UNS-1(LL-1K+) FMS | | Engineering Order 20031552-EO-01 Issue 2, dated 20.04.2006 or later approved RevisionsAircraft Flight Manual Supplement2031552-AM-01 Issue 2, dated April 2006 or later approved Revisions | Certification Basis of the Change: CS 25 Amendment 1, additionally JAA Leaflet No. 10 (Precision RNAV), AMC 20-4 (BasicRNAV), AMC 20-5 (GPS) | 27.06.2006 |
| 10014550 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02643 | | | CESSNA 501, CITIATION CESSNA 551, CITIATIONII | Installation of Dual Garmin GNS 430A/530A and 3rd COMM in Cessna Citation I model 501 and Citation II model 551. | | # Engineering Order 2027496-EO-01, issue 3, dated 30 May2007, or later approved revision.# AFM Supplement Doc. 2027496-01A, issue 2, dated 31 May 2007, or later approved revision. | 1. Certification basis of the Change CS-23, additionally:2. AMC 20-4 (Basic RNAV), AMC 20-5 (GPS), AMC 25-11 (Electronic Display Systems) and JAA TGL 7, rev.1 (VHF Com 8.33 KHz)3. Prior to installation of this modification the installer mu (...) | 20.06.2007 |
| 10014551 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02644 | | | CESSNA 500, CITATION CESSNA 550 CITATION II CESSNA S550 (CITATION II) | Installation of Dual Garmin GNS 430A/530A and 3rd COMM in Cessna Citation I model 500 and Citation II model 550, S550. | | # Engineering Order 2027496-EO-01, issue 3, dated 30 May2007, or later approved revision.# AFM Supplement Doc. 2027496-01A, issue 2, dated 31 May 2007, or later approved revision. | 1. Certification Basis of the Change CS-25 Amndt. 2, additionally:2. AMC 20-4 (Basic RNAV), AMC 20-5 (GPS), AMC 25-11 (Electronic Display Systems) and JAA TGL 7, rev.1 (VHF Com 8.33 KHz)3. Prior to installation of this modificationthe instal (...) | 20.06.2007 |
| 10014556 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02654 | | | | Honeywell EGPWS MK VI Installation | | Engineering Order 2033234-EO, Issue 2, dated 16 May 2006 or later approved revisionAFRM Supplement Doc. 2033234-AM, issue 1, May 2006 or later approved revision | Certification Basis of the Change CS 25, additionally JAA TGL 12 (TAWS) | 06.06.2006 |
| 10014559 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02658 | | | | Trimble TNL-2101 I/O Approach + GPS System Installation | | Engineering Order 2033235-EO, Issue 2, dated 15 May 2006 or later approved revisionAFM Supplemental Doc. 2033235-AM, Issue 2, May 2006 or later approved revision | Certification Basis of the Change CS 25, additionally EASA AMC 20-4 (BRNAV) and AMC 20-5 (GPS) | 06.06.2006 |
| 10014582 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02688 | | | BAE 125 -800 SERIES HAWKER 800 HAWKER 800 XP | Installation of an In-flight Entertainment System | | Engineering Order, Doc. No. 2031919-EO-01, issue 2, dated 16-May-2006, or later approved issueEASA approved Aircraft Flight Manual Supplement, Doc. No. 2031919-AM-01, Issue 3, dated 08-Feb-2007, or later approved IssueInstructions for Continued Airw (...) | This change is limited to airplanes equipped with ARINC 429Interface to GPS / FMS | 22.03.2007 |
| 10014592 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02703 | | | BAE JETSTREAM 3100 BAE JETSTREAM 3200 | Sandel ST3400 TAWS installation | | Engineering Order 2034126-EO, Issue 2, dated 12 July 2006, or later approved revisionAFM Supplement Doc. 2034126-AM, issue 1, July 2006, or later approved revision | Certification Basis of the Change CS 25, additionally JAA TGL No. 12 (TAWS) | 21.07.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|----------------|----------------|----------------------|--|---|---------------------|---|---|---------------|
| 10014605 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02728 | | | CESSNA 650 CITATION III CESSNA 650 CITATION VI | Installation of Honeywell Multi-Function Radar Display (MFRD) and as an option a Honeywell Primus 400 Radar system. | | Installation in accordance with Scandinavian Avionics Engineering Order ref 2034305-EO Issue 2, dated 29.6.2006, or later approved revisions. Operation in accordance with Scandinavian Avionics Flight Manual Supplement ref 2034305-AM Issue 1, dated Au (...) | This STC is only applicable to Aircraft without EFIS system and integrated MFD. | 11.09.2006 |
| 10014622 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02756 | | | CESSNA 500, CITATION CESSNA 550 CITATION II CESSNA 560 (CITATION V) CESSNA S550 (CITATION II) | Installation of a #Flight Display Systems# Entertainment System: DVD player, Map computer and 2 LCD displays | | System Description 2029710-DE, Drawing List 2029710-DL, MoC list 2029710-MOC, Engineering Order 2029710-EO, AFM Supplement 2029710-AM | The IFE sound shall be overridden by the PA system | 31.08.2006 |
| 10014632 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02774 | | | BEECH 60 BEECH A60 BEECH B60 | Installation of Mode S Transponder Garmin GTX 330D | | Engineering Order 2034278-EO, Issue 2, dated 14 July 2006, or later approved revision. Engineering Supplement 2034278-AM, issue 1, dated 6 September 2006, or later approved revision. Continued Airworthiness 2034278-CA, issue 1, dated 22 June 2006, or later approved revision. | | 11.10.2007 |
| 10014641 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02791 | | | CESSNA 550 BRAVO CESSNA 560 | Honeywell EGPWS MK VIII Installation | | Engineering Order 2035111-EO Issue 1, dated 16/08/2006 or later approved revision. Aircraft Flight Manual Supplement 2035111-AM Issue 1, dated August 2006 or later approved revision. | Certification Basis of the Change: CS 25 Amendment 1m, additionally JAA Leaflet No. 12 (TAWS) | 03.11.2006 |
| 10014668 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02826 | | | | Honeywell EGPWS MK VI Installation. Installation of Honeywell Enhanced Ground Proximity Warning System EGPWS/TAWS. | | # AFM Supplement Doc. 2035430-AM, Issue 2, or later EASA Approved Issue# Description of Change: 2035430-DE, Issue 1# Engineering Order: 2035430-EO, Issue 2# Technical Order: 2035430-TO, Issue 1 | 1. This STC is only applicable for aircraft which have all the following equipment already installed: Honeywell (Bendix-King) IN-182 or KMD-850 Radar Display; KN-53 NAV System; KRA-405 Radar Altimeter; KCS-55A Compass System; Computer Instruments C (...) | 30.01.2007 |
| 10014672 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02832 | | | BAE 125 -800 SERIES HAWKER 800 HAWKER HS 125-600 SERIES HAWKER HS 125-700 SERIES | Installation of a Universal Terrain Awareness Warning System (TAWS). | | Technical Order, Doc. No. 2035674-TO, issue 3, dated 03-Jan-2007, or later approved issue. EASA approved Aircraft Flight Manual Supplement, Doc. No. 2035674-AM-A, Issue 1, dated September 2006, or later approved issue for operation with FMS display (...) | 1. This change is limited to following configuration of the aircraft / equipment type: all airplanes equipped with " ARINC 429 Air Data Computers with temperature compensation, and" Rockwell Collins Proline 2 or 4 RADALT, VOR/ILS, and" Display System such as Rockwell Collins EFIS 85 | 10.01.2007 |
| 10014673 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02833 | | | BAE 125 -800 SERIES HAWKER 800 HAWKER HS 125-600 SERIES HAWKER HS 125-700 SERIES | Installation of a Universal UNS-1F and / or UNS-1L Flight Management System for Basic RNAV, Precision RNAV Operations and Non-precision Approaches and also as a primary means of navigation for oceanic/remote operations. | | Technical Order, Doc. No. 2035499-TO, Issue 2, dated 04-Jan-2007, or later approved issue. EASA approved Aircraft Flight Manual Supplement, Doc. No. 2035499-AM-A, Issue 1, dated October 2006, or later approved issue, for UNS-1F and UNS-1L FMSEASA appr (...) | 1. This change is limited to following configuration of the aircraft / equipment type: all airplanes equipped with " Rockwell Collins Proline 1 (only NAV receiver), Proline 2 or 4 Navigation Equipment, and" Rockwell Collins analog HSI#s or EFIS (...) | 10.01.2007 |
| 10014689 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02852 | | | BOEING 707-300 BOEING 737-400 BOEING 737-500 | Transponder S Mode Elementary and Enhanced Surveillance / Upgrade of existing dual diversity Transponder Mode S Installation | | Engineering Order 2035168-EO Issue 2, dated 13.12.2006 or later approved revision. Declaration of Compliance 2035168-JS, dated 20.12.2006. Aircraft Flight Manual Supplement 2035168-JS-AM Issue 1, dated December 2006 or later approved revision. | Certification Basis of the change: CS 25 Amrt. 2, additionally JAA TGL 13 Rev. 1 (Mode S ELS), JAA TGL 18 (Mode S EHS, compliance to JAA NPA 20-12a shown voluntarily. This STC is approved only for the product configuration as defined in the approved d (...) | 03.01.2007 |
| 10014690 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02853 | | | CESSNA 650 | Honeywell EGPWS MK VIII Installation | | Engineering Order 2035914-EO Issue 2, dated 10-01-2007 or later approved revisions. Aircraft Flight Manual Supplement 2035914-AM Issue 1, dated January 2007 or later approved revisions. | Certification Basis of the Change: CS 25 Amendment 2, additionally JAA Leaflet No. 12 (TAWS). | 13.02.2007 |
| 10014695 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02861 | | | BAE 125 -800 SERIES HAWKER 800 | Installation of a Honeywell Class A Enhanced Ground Proximity Warning System EGPWS/TWAS for aircrafts equipped with a compatible MFD or weather radar display. | | # Technical Order, Doc. No. 2036070-TO, Issue 1, dated 13-Dec-2006, or later approved issue. # EASA approved Aircraft Flight Manual Supplement, Doc. No. 2036070-AM, Issue 2, dated December 2006, or later approved issue. # Instructions for Con (...) | 1. This change is limited to following configuration of the aircraft / equipment type: " Rockwell Collins EFIS 85/86 with ARINC 453 interface, and" Airdata system with ARINC 429 interface (TSO-C10b or C10c), and" Rockwell Collins VIR-432 with (...) | 03.01.2007 |
| 10014702 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02872 | REV. 2 | | HAWKER 800 HAWKER 800 XP HAWKER HS 125-700 SERIES | Mode S Transponder Upgrade for existing Collins transponders system. | | Upgrade of existing Collins transponder system in accordance with Document List 2036079 DOC and Engineering Order 2036079-EO Operation in accordance with Flight Manual Suppl. 2036079-AM-A Flight Manual Suppl. 2036079-AM-B Continuing Airworthiness 2036079-CA. | 1. This STC approves only the installation of the parts contained in the above section. #Description of Type Design Change#. This STC grants no operational approval to this installation, that will need to be obtained from the State of registry for th (...) | 06.03.2007 |
| 10014703 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02874 | | | FAIRCHILD SWEARINGEN SA 228-TC FAIRCHILD SWEARINGEN SA 227-AC FAIRCHILD SWEARINGEN SA 227-BC FAIRCHILD SWEARINGEN SA 227-BC | Installation of a Garmin GPS 400 System with Remote Annunciator Switch Panel in Fairchild Swearingen SA226-TC, SA227-AC, SA227-BC | | Technical Order, Doc. No. 2036263-TO, Issue 3 or later approved issue. EASA approved Aircraft Flight Manual Supplement, Doc. No. 2036263-AM, Issue 2, dated December 2006, or later approved issue. Instructions for Continued Airworthiness, Doc. No. 203626 (...) | This change is limited to following configuration of the aircraft / equipment type: Airdata system must be either ARINC429, RS232 or Grey code as defined in the GPS 400 installation manual, and- any model of Rockwell Collins VIR-30/31/32 VOR/ILS | 14.12.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|----------------|----------------|----------------------|---|--|---------------------|--|---|---------------|
| 10014729 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02907 | REV. 1 | | CESSNA 500, CITATION CESSNA 550 BRAVO CESSNA 550 CITATION II CESSNA 560 (CITATION V) | Mode-S upgrade for existing Honeywell MST 67A transponder system. | | Means of Compliance list 2036715 MOC.AFM Supplements -2036715-AM A for aircraft without Primus 1000 EFIS;2036715-AM B for aircraft with Primus 1000 EFIS;2036715-AM C for aircraft with Primus 1000 EFIS System Description 2036715-DE;Engineering Order (...) | N/A | 03.04.2007 |
| 10014759 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02948 | | | CESSNA 650 CITATION III CESSNA 650 CITATION VI CESSNA 650 CITATION VII | Upgrade of existing MST-67A Transponder System | | Installation in accordance with Scandinavian Avionics Engineering Order ref 2036080-EO Issue 1, dated 22 December 2006, or later approved revisions.Aircraft to be operated in accordance with Scandinavian Avionics Flight Manual Supplement ref 2036080- (...) | As defined in Scandinavian Avionics Modification ref 2036080. | 12.02.2007 |
| 10014763 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02953 | | | CESSNA 425 | Replacement of one existing Transponder System by a Garmin GTX 330D Mode-S Transponder System | | Engineering Order, Doc. No. 2037527-EO, Issue 3, dated 04-May-2007 or later approved issueEASA approved Aircraft FlightManual Supplement, Doc. No. 2037527-AM-A, Issue 1, dated 13-Apr-2007 or later approved issueand, if EHS capability is given: "appr (...) | This change is limited to the configuration of aircraft / equipment type as listed under Item 4. in Engineering Order, Doc. No. 2037527-EO, Issue 3, dated 04-May-2007 or later approved issuePrior to installation of this modification the installer mu (...) | 11.05.2007 |
| 10014767 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02959 | | | CESSNA 500, CITATION CESSNA 550 CITATION II CESSNA 550, (CITATION S/II) | Installation of Honeywell Multi-Function Radar (MFRD) and an optional Honeywell Primus Radar System, if the existing radar is incompatible with the MFRD. | | Compliance Record 2036716-EO, Issue 2AFM Supplement 2036716-AM, Issue 2Instructions for Continued Airworthiness 2036716-CA, Issue 1or later approved versions. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 16.02.2009 |
| 10014777 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02975 | | | CESSNA 650 CITATION III CESSNA 650 CITATION VI CESSNA 650 CITATION VII | Installation of Honeywell TRA-45A Terrain Display/ Radio Altimeter Indicator. | | Installation in accordance with Scandinavian Avionics Engineering Order Reference 2037690-EO Issue 2, dated 20 April 2007, or later approved revisions.Aircraft to be operated in accordance with Scandinavian Avionics Flight Manual SupplementReference (...) | Aircraft Serial No. 650-0037 with FOCA STC 234-44-05 installed as defined in Scandinavian Avionics Modification Reference 2037690. | 29.05.2007 |
| 10014778 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.02976 | | | CESSNA 650 CITATION III CESSNA 650 CITATION VI CESSNA 650 CITATION VII | Installation of Sandel ST3400 TAWS System. | | Engineering Order, Doc. No. 2037448-EO, Issue 2, dated 07-March-2007 or later approved issueEASA approved Aircraft Flight Manual Supplement, Doc. No. 2037448-AM- (), Issue 2, dated 28 March 2007, or later approved revisions.STC Document List, Doc. (...) | | 04.04.2007 |
| 10014800 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03003 | | | SAAB 340B SAAB SF340A | Installation of Universal UNS-1L/K+, FMS-system. | | AFM Supplement 2037470-AM, Issue 2, 19/03/2007.Continued Airworthiness 2037470-CA, Issue 1, 26/01/2007.Engineering Order 2037470-EO, Issue 2, 19/02/2007. | N/A | 13.04.2007 |
| 10014807 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03011 | | | CESSNA 501, CITATION CESSNA 551, CITATIONII | Installation of a Honeywell CAS 67A ACAS II System and a Dual MST 67A Mode S Transponder System | | Engineering Order, Doc. No. 2038224-EO, Issue 2, dated 21-Mar-2007 or later approved issueEASA approved Aircraft FlightManual Supplement, Doc. No. 2038224-AM-A, Issue 1, dated 16-Mar-2007 or later approved issueand one of the following: " EASA approved Aircraft Flight Manual Supplement, Doc. No. | This change is limited to the configuration of aircraft / equipment type as listed under Item 4. in Engineering Order, Doc. No. 2038224-EO, Issue 2, dated 21-Mar-2007 or later approved issue. | 03.04.2007 |
| 10014809 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03013 | | | BEECH 200 BEECH 200C BEECH 300 BEECH 300LW BEECH B300 BEECH B300C | Installation of Universal UNS-1L (-1K+) FMS | | Engineering Order, Doc. No. 2037417-EO, Issue 2, dated 09-Feb-2007 or later approved issueInstallation Document List (Certification Documents), Doc. No. 2037417 STC, Issue 2, dated 12-Feb-2007, or later approved issueEASA approved Aircraft Flight M (...) | This change is limited to the configuration of aircraft / equipment type as listed under Item 4. in Engineering Order, Doc. No. 2037417-EO, Issue 2, dated 09-Feb-2007 or later approved issue. | 13.02.2007 |
| 10014823 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03036 | | | CESSNA 500, CITATION | Installation of a single diversity Mode-S transponderGarminGTX 330D and a new top antenna. | | STC Document List 2036635 STC Compliance Record AMC 2036635-CRAFAM Supplement ELS 2036635-AM-AAFM SupplementEHS 2036635-AM-B | According to EUROCONTROL definition this equipment with approval EASA.A.S.03036 is non-EHS compliant.Prior to installation of this modification the installer must determine that the interrelationship between this modification and any otherpreviously (...) | 06.06.2007 |
| 10014879 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03107 | REV. 1 | | CESSNA 650 CITATION III CESSNA 650 CITATION VI CESSNA 650 CITATION VII | Mode S Upgrade of existing Honeywell Transponder System RCZ-833 to comply with ICAO requirements for Mode S ELS and EHS;Minor change to EASA.A.S.03107;AFM Supplements for different Downlinked Aircraft Parameters.Mode S Upgrade of existing Honeywell (...) | | Installation in accordance with Scandinavian Avionics Engineering Order ref 2037689-EO Issue 2, dated 2, May 2007, or later approved revisions.Aircraft to be operated in accordance with Scandinavian Avionics Flight Manual Supplement ref 2037689-AM-A (...) | As defined in Scandinavian Avionics Modification ref 2037689.As defined in Scandinavian Avionics Modification ref 2037689. | 20.11.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|----------------|----------------|----------------------|---|---|---------------------|--|---|---------------|
| 10014910 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03154 | | | LEARJET 31 LEARJET 31A LEARJET 35 LEARJET 35A LEARJET 36 LEARJET 36A | Installation of a SANDEL ST3400 TAWS System (Class A) | | See Scandinavian Avionics:# 2040484 STC (STC Document List) Issue 1, dated 12.2.2008# 2040484-EO Engineering Order Issue 2, dated 30.08.2007# 2040484-AM AFM Supplement Issue 2, dtd. 2.10.2007# 2040484-CA Instructions for Continued Airworthiness Issue 1, dtd.18.06.2007 | Previous required aircraft configuration:All models equipped with:# Compatible Air Data Computers with ARINC 429 interface, and# Analogue Radio Altimeter interface (except ARINC 552) such asRockwell Collins ALT-50/50A , ALT-55 or HoneywellRT-200/300 (...) | 22.02.2008 |
| 10014957 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03227 | REV. 1 | | DC-9-10 SERIES DC-9-20 SERIES DC-9-30 SERIES DC-9-40 SERIES DC-9-50 SERIES DC-9-81 (MD-81) DC-9-82 DC-9-83 DC-9-87 (MD-87) MD-88 MD-90 | Mode S elementary and enhanced Upgrade of existing dual transponder system. | | Certification documents No. 2040410 DOC Issue 1 dated 27 June 2008AFMS No. 2040410-AM dated 20 February 2008Continued Airworthiness No. 2040410-CA Issue 1 dated 26 June 2008 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 24.07.2008 |
| 10014965 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03238 | | | CESSNA 501 CESSNA 551 | Dual installation of Transponder Mode S System, MST 67A with PS 550 or PS 578A Mode S Control Panel, with ELS and EHS capability. | | EASA approved Engineering Order, Doc. No. 2041564-EO, Issue2, dated 13-Aug-2007 or later approved issueEASA approved Aircraft Flight Manual Supplement, Doc. No. 2041564-AM-A, Issue 1, dated 13-Aug-2007 or later approved issue for installation with P (...) | This change is limited to the configuration of aircraft / equipment type as listed under Item 4, in EASA approved Engineering Order, Doc. No. 2041564-EO, Issue 2, dated 13-Aug-2007 or later approved issue | 26.09.2007 |
| 10015008 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03306 | | | CESSNA 500, CITATION CESSNA 550 CITATION II CESSNA S550 (CITATION II) | Installation of Honeywell GNS-XLS FMS and L3 Standby horizon Reference System. | | Master Document List 2040837 DOC, Issue 1, dated 25/11/2007, or later approved revisions. | | 28.11.2007 |
| 10015052 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03373 | | | DE HAVILLAND CANADA DHC6-300 DE HAVILLAND DHC6-310 | Dual Directional Gyro, Dual EHS/RMI and audio System. | | STG Document List 2043963 STCEngineering order and certification program. 2043963-FOAFM 204-3963-AM | As detailed in the AFM 2043963-AM | 05.03.2008 |
| 10015057 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03379 | | | BEECH 200 BEECH 200C BEECH 300 BEECH 300LW BEECH B200 BEECH B200C BEECH B300 BEECH B300C | Installation of Garmin GXW 68 / Garmin GMX 200 Weather Radar System. | | EASA approved Engineering Order, Doc. No. 2038539-EO, Issue3, dated 24-Jan-2008, or later approved IssueEASA approved Airplane Flight Manual Supplement, Doc. No. 2038539-AM, Issue 1, dated 24-Jan-2008, or later approved IssueInstructions for Continu (...) | This change is limited to the configuration of aircraft / equipment type as listed under Item 4, of EASA approved Engineering Order, Doc. No. 2038539-EO, Issue 3, dated 24-Jan-2008 or later approved issue. The system is not approved for (...) | 30.01.2008 |
| 10015061 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03384 | | | BEECH 200 BEECH 200C BEECH 300 BEECH 300LW BEECH B200 BEECH B200C BEECH B300 BEECH B300C | Installation of Garmin GPS 500 System (including compliancewith AMC 20-4, AMC 20-5 and FAA AC20-138A) | | EASA approved Engineering Order, Doc. No. 2038540-EO, Issue2, dated 22-Jan-2008, or later approved IssueEASA approved Airplane Flight Manual Supplement, Doc. No. 2038540-AM, Issue 1, dated 03-Dec-2007, or later approved IssueInstructions for Continu (...) | This change is limited to the configuration of aircraft / equipment type as listed under Item 4, of EASA approved Engineering Order, Doc. No. 2038540-EO, Issue 2, dated 22-Jan-2008 or later approved issue. The system is not approved for (...) | 30.01.2008 |
| 10015063 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03389 | | | | Sandel ST3400 TAWS Installation. | | Installed in accordance with 2044235-EO Engineering Order Issue 2 or later approved revision. Operated in accordance with 2044235-AFM AFM supplement Issue 1 or later approved revision. Maintained in accordance with 2044235-CAInstructions forContinued (...) | | 30.05.2008 |
| 10015067 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03393 | | | CESSNA 500, CITATION CESSNA 550 CITATION II CESSNA S550, (CITATION S/II) | Installation of Honeywell ART2100/KMD850 Weather Radar System. | | 1. AFM-Supplement: 2044352-AM Issue 1;2. Engineering Order: 2044352-EO Issue 2;3. Continued Airworthiness: 2044352-CA Issue 1. | 1. This STC is exclusively applicable for aircraft which have the equipment listed in chapter 4 of the Scandinavian Avionics Engineering Order 2044352-EO Issue 2 already installed;2. Prior to installation of this modification the installer must (...) | 12.03.2008 |
| 10015089 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03424 | | | FALCON 10 | Dual TDR-94D Transponder Upgrade to provide limited Enhanced Surveillance Transponder Mode S capability (Non EHS Capable). | | See Scandinavian Avionics:# 2044753 DOC (Certification Documents) issue 1, dated 12-Feb-08.# 2044753-TO TechnicalOrder (for installation instructions) issue 2, dated 11-Feb-08.# 2044753-AM AFM Supplement issue 1, dated 18-Feb -08.# 20447 (...) | Previous required aircraft configuration:# Aircraft equipped with Dual TDR-94D (-007/-008) and installation compliant with Elementary Mode S requisites.# Mode S Transponder Antenna Diversity# Collins TCAS II System# Dual ADCs installed (...) | 22.02.2008 |
| 10015113 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03460 | | | FAN JET FALCON SERIES C FAN JET FALCON SERIES D FAN JET FALCON SERIES E FAN JET FALCON SERIES F FAN JET FALCON SERIES G MYSTERE FALCON 20-C5 MYSTERE FALCON 20-D5 MYSTERE FALCON 20-E5 MYSTERE FALCON 20-F5 MYSTERE FALCON 200 | Dual TDR-94D Transponder Installation Upgrade to provide both Elementary and Enhanced Surveillance Transponder Mode S Capabilities. | | See Scandinavian Avionics:# 2045142-STC (Certification Documents) issue 2, dated 04 Jun 08.# 2045142-EO Engineering Order issue 2, dated 11 Feb 08.# 2045142-AM-A AFM Supplement for Mode S Transponder Elementary Surveillance Capability, issu (...) | Previous required aircraft configuration:# Aircraft equipped with Dual TDR-94D installation with Mode S Diversity Antenna installation and Radio Tuning Unit RTU-4220# ADCs with ARINC 429 Interface.# Collins TCAS II System# Collins Prol (...) | 09.06.2008 |
| 10015128 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03479 | | | BOMBARDIER DHC-6 SERIES 300 | Honeywell EGWWS MK VI, TRA-45 Display installation. | | AFM2045504-AM Engineering Order 2045504-EO. | | 28.03.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|------------------|----------------|----------------------|---|--|--|---|--|---------------|
| 10015131 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03482 | | | BEECH 300 BEECH 300LW BEECH B300 BEECH B300C | Installation of an L3 F-1000 Digital Flight Data Recorder System | | EASA approved Engineering Order, Doc. No. 2045502-EO, Issue2, dated 24-Apr-2008, or later approved IssueEASA approved Airplane Flight Manual Supplement, Doc. No. 2045502-AM, Issue 2, dated 14-Apr-2008, or later approved IssueInstructions for Continu (...) | This change is limited to the configuration of aircraft / equipment type as listed under Item 4. of EASA approved Engineering Order, Doc. No. 2045502-EO, Issue 2, dated 24-Apr-2008 or later approved issue. | 05.05.2008 |
| 10015132 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03483 | | | BEECH 200 BEECH 200C BEECH 300 BEECH 300LW BEECH B200 BEECH B200C BEECH B300 BEECH B300C | Installation of a Universal CVR-30B or CVR-120 Cockpit Voice Recorder | | EASA approved Engineering Order, Doc. No. 2045503-EO, Issue2, dated 23-Apr-2008, or later approved IssueEASA approved Airplane Flight Manual Supplement, Doc. No. 2045503-AM, Issue 2, dated 14-Apr-2008, or later approved IssueInstructions for Continu (...) | This change is limited to the configuration of aircraft / equipment type as listed under Item 4. of EASA approved Engineering Order, Doc. No. 2045503-EO, Issue 2, dated 23-Apr-2008 or later approved issue. | 05.05.2008 |
| 10015137 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03490 | | | CESSNA 650 CITATION VII | UPGRADE of existing FMS Universal UNS-1Csp to UNS-1Csp+ to comply with P-RNAV requirements. | | 2045402-EO Engineering Order Issue 2 or later approved revision.2045402-AM AFM Supplement Issue 1 or later approved revision.2045402-CA Instructions for Continued Airworthiness Issue 1 or later approved revision. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 25.08.2008 |
| 10015170 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03544 | | | BEECH 400 BEECH 400A | Mode S ELS/EHS Upgrade of existing transponder system. | | Engineering order 2042065-EO issue 2.AFMS 2042065-AM issue 1. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 10.06.2008 |
| 10015183 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03563 | | | | Replacement of existing COM 1 / 2 and NAV 1 / 2 systems by Garmin GNS 530 (GNS 1) and Garmin GNS 430 (GNS 2) | | EASA approved Engineering Order, Doc. No. 2046711-EO, Issue1, dated 21-Apr-2008, or later approved IssueEASA approved Airplane Flight Manual Supplement, Doc. No. 2046711-AM, Issue 1, dated 21-Apr-2008, or later approved IssueInstructions for Continu (...) | This change is limited to the configuration of aircraft / equipment type as listed under Item 4. of EASA approved Engineering Order, Doc. No. 2046711-EO, Issue 1, dated 21-Apr-2008 or later approved issue. | 17.06.2008 |
| 10015194 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03580 | | | BAE 125 -800 SERIES HAWKER HS 125-600 SERIES HAWKER HS 125-700 SERIES | Installation of a Digital Flight Data Recorder. | | Installed in accordance with 2047003-EO Engineering Order Issue 2 or later approved revision. Operated in accordance with 2047003-AM AFM supplement Issue 1 or later approved revision. Maintained in accordance with 2047003-CA Instructions for Continued (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the (...) | 15.06.2009 |
| 10015274 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.A.S.03769 | | | LEARJET 55 | Dual Universal UNS-1F FMS Installation. | | As noted in Certification Document List 2051373 DOC, issue 1 (or later approved revisions). | # Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. (...) | 17.04.2009 |
| 10017138 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.R.S.01237 | | | EUROCOPTER DEUTSCHLAND BO-105 | Installation of Bendix/King Radar Altimeter System KRA 405/405B | | RFM Supplement Model BO 105 series, Doc 2021602-AM-01, Issue 3, April 2006 or later EASA approved issue | | 21.04.2006 |
| 10017188 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.R.S.01325 | | | EUROCOPTER FRANCE EC 120 B | VHF COMM, XPDR, ICS, Directional Gyro and Attitude Indicator. | | 1. Installation instructions:- Technical Order for installation 2029510-TO-01 Iss. 2, dated 27/03/2006.2. Instructions for continued Airworthiness:- Continued Airworthiness 2029510-CA-01 Iss. Dated 30/09/2005. | 1. The installation of this STC is not approved for Night VFR approved aircraft.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed (...) | 09.10.2007 |
| 10017234 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.R.S.01412 | | | EUROCOPTER FRANCE EC 120 B | Installation of Avionics System Garmin GNS 480 / HoneywellKY-96A / Garmin GTX 33 as defined in Engineering Order 2042580-EO. | | - Definition and Installation in accordance with:-Scandinavian Avionics Design Engineering Order 2042580 -EO, issue 1, dated 25.01.2008.- Inspection and Maintenance in accordance with:-Scandinavian Avionics Design Instruction forContinued (...) | | 19.03.2008 |
| 10017235 | SCANDINAVIAN AVIONICS DESIGN ApS | EASA.R.S.01413 | | | EUROCOPTER FRANCE EC 120 B | Installation of several avionics system as defined into Engineering Order No. 2042316 # EO issue 2 | | Scandinavian Avionics Design GmbH Engineering Order No. 2042316 # EO issue 2 Scandinavian Avionics Design GmbH Drawing List No. 2042316 # DL issue 1 dated 01.02.2008. Scandinavian Avionics Design GmbH RFM Supplement ref. 2042316 - (...) | This change is limited to Rotorcraft equipped with Garmin GNS 480 Installation. | 04.04.2008 |
| 10028447 | SCANDINAVIAN AVIONICS DESIGN ApS | P-EASA.A.S.03759 | REV. 1 | | A318, A319, A320, A321 | Installation of provisions for the use of Dual navAero Electronic FlightBag Class 2 (EFB) System | In accordance with the EASA TCDSThe Certification Basis for the original product remains applicable to this certificate/approvalThe certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | - Engineering Order & Certification Program, ref. doc. 2051375-EO, Issue 1, dated 22nd December 2009- Airplane Flight Manual Supplement, ref. doc. 2051375-AM, Issue 3, dated 15th December 2009- Instruction for Continued Airworthiness, ref. doc. 205137 (...) | Not Applicable | 12.01.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|----------------|----------------|----------------------|---|---|---|---|--|---------------|
| 10033960 | SCANDINAVIAN AVIONICS DESIGN ApS | | | | B200 | Provisions for and installation of a SLAR and a Homer antenna. | The Certification Basis of the original product and the following additional /, alternative airworthiness requirements are applicable to this change For the change CS-23, Amtd. 2, has been elected to comply with where possible The certificated noise and/ (...) | EASA approved Engineering Order, Doc. No. 2063299-EO, Issue 1, dated 21-Feb-2011, or later approved issue. EASA approved Aircraft Flight Manual Supplement, Doc. No. 2063299-AMA, Issue 1, dated 16-Feb-2011 (<->) or later approved issue. Approved Instruct (...) | Applicable for serial number BB-1936 only. | 22.02.2011 |
| 10013976 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.0097 | REV. 2 | | 550 (CITATION BRAVO) 550 (CITATION II) 560 (CITATION ULTRA) 560 (CITATION V) S550 (CITATION S/II) | Installation of an L3 Digital Flight Data Recorder Revision 2 was raised due to missing technical document references in Revision 1 | EASA TCDS IM.A.2077 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate approval JAR 25 at change 15 The requirements for environmental protection and (...) | - Master Documents List 2018778-01 DOC issue 8, dated 27 Sept. 2010- Engineering Order 2018778-EO-01 Issue 5, dated 26 Sept. 2010- Airplane Flight Manual Supplement 2018778-AM-01A, issue 5 from 19 January 2010 for 115 VAC or alternative Airplane Flig (...) | Limited to: S/N 550-0001 through 550-0733 (Citation II) S/N 550-0801 through 550-1136 (Citation Bravo) S/N 560-0001 through 560-0259 (Citation V) S/N 560-0260 through 560-0538 (Citation Ultra) S/N S550-0001 through S550-0160 (Citation S/II) | 05.10.2010 |
| 10014039 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.01112 | REV. 2 | | CESSNA 525 CESSNA 525A | Installation of dual GTX 330D Mode S Transponder systems. Installation of dual GTX 330D Mode S Transponder systems. | Design: Non-significant change The Certification Basis of the original product and the following additional /, alternative airworthiness requirements are applicable to this change For the change CS-23 has been elected to comply with where possible. Environm (...) | - EASA approved Engineering Order, Doc. No. 2019974-EO-05, Issue 4, or later approved issue- for ELS-mode operation of XPDR system: EASA approved Airplane Flight Manual Supplement, Doc. No. 2019974-AM-05A, Issue 1, or later approved issue- 1 (...) | 1. According to Engineering Order 2019974-EO-05, Issue 41. According to Engineering Order 2019974-EO-05, Issue 4 | 25.02.2008 |
| 10014148 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.01268 | REV. 1 | | 1. SA226-TC, SA227-AC 1. SA227-BC 2. A226-T, SA226-AT 2. SA226-T(B), SA227-AT 2. SA227-TT | Installation of Honeywell CAS 67A ACAS II System with dual Mode S Transponder | Design: Non-significant change The Certification Basis of the original product and the following additional /, alternative airworthiness requirements are applicable to this change For the change CS-23 has been elected to comply with where possible. Environm (...) | EASA approved Engineering Order, Doc. No. 2027046-EO, issue 2, dated 25-Oct-2010, or later approved issue EASA approved Aircraft Flight Manual Supplement, Doc. No. 2027046-AM-01A, Issue 2, dated 07-Jun-2005 (<->) or later approved issue, and EASA approve (...) | This change is limited to the configuration of aircraft / equipment types listed under Chapter 4, in EASA approved Engineering Order, Doc. No. 2027046-EO, Issue 2, dated 25-Oct-2010, or later approved issue. | 08.11.2010 |
| 10014489 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.02551 | REV. 1 | | MODEL 35, MODEL 35A, MODEL 36, MODEL 36A. | Installation of Dual Universal UNS-1E FMS in Learjet 35/35A/36/36A Major Change to existing STC by additional certification for P-RNAV operations by showing compliance with JAA TGL-10. Defined in Engineering Order 2032349-EO, issue 5, dated 18 April 20 (...) | Certification Basis for the change: CS25, Amtd. 01. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Scandinavian Avionics, AFM Supplement 2032349-AM issue 2, dated 30 March 2006 Scandinavian Avionics, AFM Supplement 2032349-AM B, issue 1, dated 23 November 2007 Scandinavian Avionics, AFM Supplement 2032349-AM C, issue 2, dated 10 April 2013 Scandinavian (...) | 1. The system must utilize software version 801.X (for X 6-9) or 802.X (for X 6-9) (ref Scandinavian Avionics, AFM Supplement 2032349-AM C, limitations section, issue 2, dated 10 April 2013) 2. Software Limitation of original AFMS (2032349-AMor 2032349 (...) | 07.05.2013 |
| 10014598 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.02718 | REV. 1 | | SUPER KING AIR 200 | Installation of Navids Flight inspection system UNIFIS 3000 Installation in Aircraft includes: Two static inverters for power supply; A vertical gyro VG400; Electrical installation with Relays and fuse panel; An advisory light and shut-off switch in the (...) | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 23 amendment 23-0 unless otherwise noted: CS23.301(a), 23.303, 23.305(a)(b), 23.307(a), 23.333(a), 23.603, 23.605(a), 23 (...) | - Air Alpha design package EO-B200-25-105-Flight Manual supplement: FMS.EO.B200-25-105-UNIFIS 3000 Installation and Maintenance Manual version 1 Aug 2006- Unifis 3000 Operator Manual version 1, Aug 2006- Supplemental Installation & Maintenance Manual f (...) | 1. Installed unit, UNIFIS 3000 must comply with requirements in JAA TGLN.17.2. Antenna installations and pilot aids as a CDI with annunciator and intercom must be approved separately. 3. This STC is approved only for the product configuration as defin (...) | 12.08.2013 |
| 10014618 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.02750 | REV. 1 | | B200 SUPER KING AIR | Antennas - Relocation and Installation in accordance with EO-B200-25-106 and DWG No. EN-B200-25-106-01 to support UNIFIS 3000 Flight Inspection System. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 23 amendment 23-0: CS23.301(a), 23.303, 23.305, 23.307, 23.335(b), 23.337, 23.365, 23.571(c), 23.603, 23.605, 23.609, 23 (...) | - Flight Manual supplement FMS.EO-B200-25-106, Rev.0, 07 June 06; - Supplemental Maintenance Manual SMM ATP25.0104, Rev.0, 07 June 06; or later revisions of the above listed documents approved by EASA; (...) | MSN BB1221 only. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft configurations shall be determined by the installer. The install (...) | 15.08.2013 |
| 10014696 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.02863 | REV. 1 | | 1. A226-T, SA226-AT 1. SA226-T(B), SA227-AT 2. SA226-TC, SA227-AC 2. SA227-BC | Installation of Sandel ST3400 TAWS System | Design: Non-significant change The Certification Basis of the original product and the following additional /, alternative airworthiness requirements are applicable to this change For the change CS-23 has been elected to comply with where possible. Environm (...) | EASA approved Engineering Order, Doc. No. 2035640-EO, issue 4, dated 25-Oct-2010, or later approved issue EASA approved Aircraft Flight Manual Supplement, Doc. No. 2035640-AM-A, Issue 4, dated 15-Jul-2010 (<->) or later approved issue, or EASA approved (...) | This change is limited to the configuration of aircraft / equipment types listed under Chapter 4, in EASA approved Engineering Order, Doc. No. 2035640-EO, Issue 4, dated 25-Oct-2010, or later approved issue. | 08.11.2010 |
| 10014798 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.03001 | REV. 1 | | BEECH B200 | 1. Installation of UNIFIS 3000 Flight Inspection System including antennas in Beech B200.2. The system is used to check and calibrate ground based navigation aids. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 23 amendment 23-0 unless otherwise noted: CS23.251, 23.301, 23.303, 23.305, 23.307, 23.365, 23.561, 23.571(c), 23.572(a) (...) | - Air Alpha Master Drawing No. EN-B200-25-116-01, Rev.1, dated 30 August 2007 Modification Front sheet for modification EO-B200-25-116 with complete listing of Declaration of Compliance, Report EO-B200-25-116; - Air Alpha Flight Manual Supplement for B (...) | Limited to B200 S/N BB-829. The installation of this modification by third persons is subject to written permission of the approval holder and disposal of the approved appropriate documentation. Prior to installation of this modification it must be de (...) | 15.08.2013 |
| 10014973 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.03259 | REV. 5 | | 737-300, -400, -500, -600 737-700, -800, -900 | Installation of a Dual navAero Electronic and logging display system. Change consisting in installing a dual portable Class 2 EFB's t.BagC2 system | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this approval: CRI H-1 dated 7/11/2011 The certificated noise and/ or emissions levels of the original product are (...) | - Engineering order and certification program 2067424-EO issue 1 dated 22/08/2011- AFM supplement 2067424-AM-A issue 2 dated 14/10/2011- AFM supplement with 3G automatic 2067424-AM-B issue 2 dated 14/10/2011- Instructions for Continuing Airworthiness (...) | Not Applicable | 14.11.2011 |
| 10014975 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.03262 | REV. 1 | | BEECH B200 | Garmin GWX-68 Radar, Garmin GMX-200 Multi-Function Display, and Avidyne TAS620 Traffic Advisory System. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 23 amendment 23-0: CS23.251, 23.301, 23.303, 23.305, 23.307, 23.341, 23.365, 23.561, 23.571(a), 23.573(b), 23.601, 23.60 (...) | Flight Manual Supplement FMS.EO-B200-34-118, Engineering Order (EO-B200-34-118) and related documents or later revisions of the above listed documents approved by EASA. | Limited to S/N BB829. The installation of this modification by third persons is subject to written permission of the approval holder and disposal of the approved appropriate documentation. Prior to installation of this modification it must be determin (...) | 15.08.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|------------------|----------------|----------------------|---|--|---|---|---|---------------|
| 10014976 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.03264 | REV. 1 | | BEECH B200 | Installation of two Sandel SN3500 EHSI (Electronic Horizontal Situation Indicator) replacing existing HSI displays. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 23 amendment 23-0; CS 23.301, 23.303, 23.305, 23.307, 23.341, 23.561, 23.601, 23.603, 23.605, 23.607, 23.611, 23.613, 23. (...) | Flight Manual Supplement FMS.EO-B200-34-119, Revision 1 and Revision 2, Engineering Order (EO-B200-34-119) and related documents or later revisions of the above listed documents approved by EASA. Note: Revision 2 of the AFMS has been approved with minor change EASA.A.A.01890. | Limited to S/N BB829. The installation of this modification by third persons is subject to written permission of the approval holder and disposal of the approved appropriate documentation. Prior to installation of this modification it must be determined (...) | 15.08.2013 |
| 10015305 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.03894 | REV. 1 | | 1. 200, 200C, 300, 300LW 2. B200, B200C, B300, B300C | Dual TDR-94D Mode S ELS/EHS Transponder System Upgrade, and optional for ADS-B | The Certification Basis of the original product and the following additional / alternative airworthiness requirements are applicable to this change. For the change CS-23, Amtd. 2, has been elected to comply with where possible. The certificated noise and/ (...) | EASA approved Engineering Order, Doc. No. 2053779-EO, issue 2, dated 21-Feb-2012, or later approved issue EASA approved Aircraft Flight Manual Supplement, Doc. No. 2053779-AM-A, issue 1, dated 27-Apr-2009 (-s, -c) or later approved issue, for ELS only EASA (...) | This change is limited to the configuration of aircraft / equipment types listed under Chapter 4, in EASA approved Engineering Order, Doc. No. 2073065-EO, Issue 1, dated 22-Mar-2012, or later approved issue for installations without ADS-B, or Issue 2 (...) | 30.03.2012 |
| 10026864 | SCANDINAVIAN AVIONICS DESIGN GMBH | P-EASA.A.S.03938 | REV. 1 | | CESSNA 500, 550 | GARMIN GPS 400W/500W installation. Revision 1 supersedes the previous issue of this STC to delete the model S550 from the applicability list for correction. | Certification Basis as per EASA TCDS IM.A.207 Issue 2 from 5 March 2009 | Master Document List "2054628-DOC" issue 1 from 17 August 2009. Aircraft Flight Manual Supplement "2054628-AM" issue 1 from 11 August 2009. Instructions for Continued Airworthiness "2054628-CA" issue 1 from 11 August 2009 or later revisions of the above listed documents approved by EASA | The limitations according to Aircraft Flight Manual Supplement 2054628-AM are applicable. | 18.08.2009 |
| 10027014 | SCANDINAVIAN AVIONICS DESIGN GMBH | P-EASA.A.S.03849 | | | 500 (CITATION AND CITATION I) | Installation of Honeywell MST 67A Dual Mode S XPDR transponder. | Certification Basis as per EASA TCDS IM.A.207 | Engineering Order 2052922-EO Issue 1 from 10 June 2009- List of Certification Documents 2052922-DOC Issue 2 from 28. August 2009- AFM Supplements 2052922-AM A Issue 1 from 03 March 2009, 2052922-AM B Issue 1 from 03 March 2009, 2052922-AM C Issue 1 (...) | None | 31.08.2009 |
| 10028068 | SCANDINAVIAN AVIONICS DESIGN GMBH | P-EASA.A.S.03732 | | | 650 (CITATION VII) | Software upgrade of installed Honeywell GNS-XL or GNX-XLS FMS system for P-RNAV operation | CS 25 amendment 4. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Engineering Order 2050529-EO Issue 2 from 27 November 2009. List of STC documents "2050529 STC" issue 2 from 27 November 2009. AFM Supplement 2050529-AM-A issue 2 from 27 November 2009 for GNS-XL FMS AFM Supplement 2050529-AM-B issue 2 from 27 November 2009 (...) | Limited to: S/N 650-7001 thru 650-7119. This STC applies only for Single or Dual FMS GNS-XL or GNS-XLS configuration, approved for B-RNAV operation. Interim with other FMS is not approved within this STC. | 30.11.2009 |
| 10028206 | SCANDINAVIAN AVIONICS DESIGN GMBH | P-EASA.R.S.01586 | | | R22 BETA | Title: Camera with support Description: Optional Installation of a camera turret, related recorder and data link transmitter system in accordance with Master Document List 2053852 DOC issue 2, dated 04.12.2009. The installation is based on Scandinavian Avionics change no. 2053852 | CS27 Amtd. 2. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | POH Supplement Robinson Model R22 Beta Doc. 2053852-AM Iss. 2. or later revisions of the above listed documents approved by EASA- Instructions for Continued Airworthiness 2053852-CA, Iss. 1 or later revision, Installation Instructions 2053852 STC Iss. 1 or later revision | S/N 0501 and subsequent, Pitot heat and Search Light must not be installed, ... | 10.12.2009 |
| 10028446 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.03971 | REV. 1 | | BEECH 200, BEECH B200 BEECH B200C, BEECH B200CGT BEECH B200GT | Installation of Blue Sky Network (BSN) D1000(A) Iridium Satellite Communication and Tracking | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23.B NSCT. The requirements for environmental protection and the associated certified noise and/ or emissions levels of th (...) | Installation in accordance with the data listed in Section 4.1 of the Air Alpha A/S Master Data List Numbered: MDL.EO-B200-23-132 Rev. 07. Operation in accordance with Air Alpha A/S Flight Manual Supplement Numbered: FMS.EO-B200-23-132 Rev. 00. Instruct (...) | The installation of this modification by third persons is subject to written permission of the approval holder and disposal of the approved appropriate documentation. Prior to installation of this modification it must be determined that the interrelat (...) | 15.08.2013 |
| 10028452 | SCANDINAVIAN AVIONICS DESIGN GMBH | P-EASA.R.S.01580 | REV. 1 | | EC 120 B | Replacement of existing SAFT Ni-Cad. battery with Concorde lead acid battery RG-350. Note: this STC is issued for the purpose of transferring STC No. 10028452 from Air Alpha A/S (DOA 21J.268) to new STC holder Scandinavian Avionics Design GmbH (DOA 21J.014) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 27.301 (a) CS27.303, CS 27.305 (a), CS 27.307(a), CS 27.561, CS27.601, CS27.603, CS (...) | Operation in accordance with:- Air Alpha A/S Flight Manual Sup. FMS.EO-EC120-24-133 Rev. 00. or later revisions of the above listed documents approved by EASA. Definition and Installation in accordance with:- Air Alpha A/S Master Data List MDL.EO-EC1 (...) | None | 12.12.2012 |
| 10028645 | SCANDINAVIAN AVIONICS DESIGN GMBH | | REV. 1 | | B200, B200C, B200GT, B200CGT | Air Alpha STC Definition Document EO-B200-23-138 for Intercom installation and modification. Intercom installation in cabin for special mission Aircraft. The cabin intercom interfaces to the cockpit intercom. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23 has been elected apart from CS-23-561 where FAR Part 23.561 Amtd 23.7 is used. The requirements for environmental protection (...) | Air Alpha Master Data List No. MDL.EO-B200-23-138 or later revisions of the above listed documents approved by EASA | The installation of this modification by third persons is subject to written permission of the approval holder and disposal of the approved appropriate documentation. Prior to installation of this modification it must be determined that the interrelat (...) | 15.08.2013 |
| 10031128 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | ATR 42-200/300/320 | Installation of Rockwell Collins TCAS-94 System with Single TDR-94D System, as per Scandinavian Avionics Engineering Order 2056096-EO issue 1 dated 27th September 2010 or later EASA Approved revisions. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: EASA CS25 amendment 4 plus:- EASA Special Condition CRI F-001 Issue 3 "HIRF Protection" (...) | See Scandinavian Avionics Design documents:- Certification Documents 2056096 DOC issue 2- Engineering Order 2056096-EO Issue 1- Aircraft Flight Manual Supplement (XPDR) 2056096-AM-A Issue 2- Aircraft Flight Manual Supplement (XPDR) 2056096-AM-B Issue (...) | None | 11.10.2010 |
| 10031822 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.R.S.01449 | | | EUROCOPTER FRANCE AS 365N EUROCOPTER FRANCE AS 365N1 EUROCOPTER FRANCE AS 365N2 EUROCOPTER FRANCE AS 365N3 | | | | | 25.06.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|------------------|----------------|----------------------|--|--|---|---|---|---------------|
| 10031899 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | HAWKER 800XP | Installation of an International Communications Group Inc. ICS-100Iridium Satellite Telephone. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-25, Am 4. The requirements for environmental protection and the associated noise and/or emissions levels of the original product are (...) | Installation in accordance with Scandinavian Avionics Design document ref 2056325 STC (STC Document List) Operation in accordance with Scandinavian Avionics Design document ref 2056325-AM (Aircraft Flight Manual Supplement) Instructions for Continued Ai (...) | Model Hawker 800XP, S/N: 258358 | 20.09.2010 |
| 10032285 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | SF340A AND 340B | Installation of Collins TCAS-94 system with dual Collins TDR-94 Transponders. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-25, am 4CRI H-01, Issue 2 The requirements for environmental protection and the associated noise and/or emissions levels of the original product are (...) | Engineering Order and Certification Program 2055179-EO, Issue 1: List of Certification Documents 2055179DOC, Issue 1: List of Installation Documents 2055179STC, Issue 1: AFM-supplement 2055179-AM-A, Issue 2: AFM-supplement 2055179-AM-B, Issue 2 or later (...) | Transponder system NOT approved for ELS or EHS. | 22.10.2010 |
| 10032445 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | SA226-T(B), SA227-AT, SA227-TT SA26-AT, SA226-T, SA226-AT | Installation of Garmin GNS 530AW / GNS 430AW as dual installation together with a Honeywell KMR 675 Marker Beacon Receiver | Design: Non-significant change The Certification Basis for the original product and the following additional /, alternative airworthiness requirements are applicable to this change For the change CS-23 has been elected to comply with where possible. Environmental (...) | EASA approved Engineering Order, Doc. No. 2061255-EO, issue 1, dated 26-Oct-2010, or later approved issue EASA approved Aircraft Flight Manual Supplement, Doc. No. 2061255-AMA, issue 1, dated 25-Oct-2010 (>, <)- or later approved issue, for GNS 430AW and G (...) | This change is limited to the configuration of aircraft / equipment types listed under Chapter 4. in EASA approved Engineering Order, Doc. No. 2061255-EO, issue 1, dated 26-Oct-2010, or later approved issue. | 08.11.2010 |
| 10033395 | SCANDINAVIAN AVIONICS DESIGN GMBH | | REV.1 | | CESSNA 500 CESSNA 550 CESSNA S550 | GPS / Navigation / Communication, Dual Garmin GNS 430/530 AW installation The installation includes GPS antenna and single marker beacon receiver (KMR 675 or 5126) installation | EASA TCDS I.M.A.2077 The Certification Basis for the original product remains applicable to this certificate/ approval The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are (...) | „Master Document List “2053445 DOC” Issue 3 from 20 Jan. 2011. „Aircraft Flight Manual Supplement for GNS 430/530 AW, VHF Communication Receiver, VOR/ILS Receiver / GPS Receiver “2053445-AM A” Issue 1 from 01 June 2010. „Aircraft Flight Manual Supple (...) | None | 21.01.2011 |
| 10033781 | SCANDINAVIAN AVIONICS DESIGN GMBH | P-EASA.A.S.03767 | | | BEECH 1900C, AIRLINER BEECH 200 BEECH 200C BEECH 300 BEECH 300LW BEECH B200 BEECH B200C BEECH B300 BEECH B300C | Installation of Sandel SA4550 EADI and Sandel SA4500 EHSI Collins AHS-1000 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. For the change CS-23 has been elected to comply with The certificated noise and/ or emi (...) | EASA approved Engineering Order, Doc. No. 2050691-EO, issue 1, dated 25-May-2009, or later approved issue- EASA approved Aircraft Flight Manual Supplement, Doc. No. 2050691-AM, Issue 1, dated 01-Jun-2009 or later approved issue- Approved Instruction (...) | This change is limited to the configuration of aircraft / equipment types listed under Item 4. in EASA approved Engineering Order, Doc. No. 2050691-EO, issue 1, dated 25-May-2009 or later approved issue. | 01.07.2009 |
| 10033782 | SCANDINAVIAN AVIONICS DESIGN GMBH | P-EASA.A.S.03768 | | | 200, 200C, B200 B200C, 300, 300LW, B300 B300C, 1900C | Installation of Garmin GNS 430AW/GNS 530AW | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. For the change CS-23 has been elected to comply with. The certificated noise and/ or em (...) | EASA approved Engineering Order, Doc. No. 2051374-EO, issue 1, dated 10-Jun-2009, or later approved issue- EASA approved Aircraft Flight Manual Supplement, Doc. No. 2051374-AMA, Issue 1, dated 14-May-2009 or later approved issue, together with- EASA (...) | This change is limited to the configuration of aircraft / equipment types listed under Item 4. in EASA approved Engineering Order, Doc. No. 2051374-EO, issue 1, dated 10-Jun-2009 or later approved issue. | 01.07.2009 |
| 10034415 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.03858 | | | BEECH 400, 400A | Installation of various Entertainment Systems acc.: Engineering Order Doc.: 2051195-EO Compliance Record Doc.: 2051195-COR | CS-25 | -AFM Supplement Doc.: 2051195-AM- Instructions for Continuous Airworthiness Doc.: 2051195-CA | None | 28.07.2009 |
| 10036450 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | GULFSTREAM 200 / GALAXY | UPGRADE OF ROCKWELL COLLINS TDR-94D MODE S TRANSPONDER SYSTEM FORENHANCED (EHS) SURVEILLANCE CAPABILITY. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval Cert base as TCDSJAR 25 Change 13 Applicant elected to comply CS 25, Amdt. 5, AMC 20-13 a (...) | To be installed in accordance with Engineering Order Doc. ID 2062013-EO, Issue 1 and Technical Order Doc. ID 2062013 Issue 1 To be operated in accordance with Aeroplane Flight Manual Supplement Doc. ID 206213-AM, Issue 1 To be maintained in accordance C (...) | None | 09.09.2011 |
| 10036673 | SCANDINAVIAN AVIONICS DESIGN GMBH | P-EASA.A.S.03703 | | | SD3-30, SD3-60 SD3-SHERPA, SD3-60 SHERPA | Installation of GARMIN GPS-400 | CS 25 Amendment 4 The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Scandinavian Avionics Engineering Order 2042759-EO Issue 1 dated 21 July 2011 Flight Manual Supplements 2042759-AM-A Issue 2, dated 15 March 2011 2042759-AM-C Issue 2, dated 15 March 2011 Continued Airworthiness 2042759-CA Issue 2, dated 10 March 2011 or la (...) | None | 28.09.2011 |
| 10036958 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | BAE 146, AVRO 146-RJ | Provision for a Dual NAVAERO 1.BagC2 Class 2 Electronic Flight Bag System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. EASA Certification Specification CS-25 Amendment 4CRI H-01 Instructions for Continued (...) | Scandinavian Avionics Engineering Order and Certification Program Provision for a Dual NAVAERO 1.BagC2 Class 2 EFB System in a BAE 146 / Avro 146-RJ Series, 2064098-EO Issue 2 dated 18 October 2011 Scandinavian Avionics Certification Documents, 206409 (...) | None | 19.10.2011 |
| 10037345 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | CESSNA 525B | Installation of L3 FA2100 Digital FDR | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | According to Engineering Order No. 2068970-EO Issue 1 To be operated in accordance with AFM-S Document No. 2068970-AM Issue 1 To be maintained i.a.w. CA Document No. 2068970-CA Issue 1 or later revisions of the above listed documents approved by EASA | S/N 525B-135 and on According limitation section of AFM-S | 18.11.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|----------------|----------------|----------------------|---|--|---|--|--|---------------|
| 10038456 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | ATP | Installation of a Five Panel Universal EFI-890R EFIS. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: EASA Certification Specification CS-25 Amendment 6C RI A-1 EASA Supplemental Type Certif (...) | Scandinavian Avionics Engineering Order and Certification Program, 5panel EFIS Universal EFI-890R Installation, 2051192-EO Issue 3 dated 15 February 2012; Scandinavian Avionics Certification Documents, 2051192-DOC Issue 1 dated 15 February 2012; Scandina (...) | None | 17.02.2012 |
| 10039000 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | 1. 200, 200C, 300, 300LW 2. B200, B200C, B300, B300C | Installation of Universal UNS FMS System | The Certification Basis of the original product and the following additional / alternative airworthiness requirements are applicable to this change. For the change CS-23, Amdt. 2, has been elected to comply with where possible. The certificated noise and/ (...) | EASA approved Engineering Order, Doc. No. 2073065-EO, Issue 1, dated 22-Mar-2012, or later approved issue. EASA approved Aircraft Flight Manual Supplement, Doc. No. 2073065-AM A, Issue 1, dated 06-Mar-2012 <-,-,-> or later approved issue. Approved Instruct (...) | This change is limited to the configuration of aircraft / equipment types listed under Chapter 4. in EASA approved Engineering Order, Doc. No. 2073065-EO, Issue 1, dated 22-Mar-2012, or later approved issue. | 29.03.2012 |
| 10040003 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | B300, B300C | Installation of Mission Antennas and a Drop Tube | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-23, Amendment 2 (not required by 21A.101) The requirements for environmental protecti (...) | AFM Supplement for Drop Tube 2069789-AFM, Issue 1 or later revisions of the above listed documents approved by EASA. Continued Airworthiness 2069789-CA, Issue 1. Master Document List 2069789-DOC, Issue 1. Engineering Order 2069789-EO, Issue 1 | none | 08.06.2012 |
| 10040472 | SCANDINAVIAN AVIONICS DESIGN GMBH | | REV. 1 | | MU-300 (DIAMOND I) MU-300 (DIAMOND IA) | Digital FDR Installation | The Certification Basis of the original product and the following additional / alternative airworthiness requirements are applicable to this change. For the change CS-25, Amdt. 4, has been elected to comply with where possible. The certificated noise and/ o (...) | Installation in accordance with EASA approved Engineering Order, Doc. No. 2076242-EO, Issue 1, dated 26-June-2012, or later approved issue. Operation in accordance with EASA approved Aircraft Flight Manual Supplement, Doc. No. 2076242-AM, Issue 1, dat (...) | This change is limited to the configuration of aircraft / equipment types listed under Chapter 4. in EASA approved Engineering Order, Doc. No. 2076242-EO, Issue 1, dated 26-June-2012, or later approved issue. | 09.07.2012 |
| 10040973 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | ATP | FMS Upgrade for P-RNAV | CS 25 Amendment 4 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The requirements for environmental protection and the associated certi (...) | Scandinavian Avionics Engineering Order 2071338-EO Issue 1 dated 05 Jul 12 Flight Manual Supplement 2071338-AM Issue 1 dated 29 May 12 Continued Airworthiness 2071338-CA Issue 1 dated 05 Dec 11 or later revisions of the above listed documents approved by EASA | none | 10.08.2012 |
| 10042855 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | 228-200/201-/202-/212 | Installation of GARMIN GTN750 / GTN650 and Sandel SN4500 EHSI | The Certification Basis of the original product and the following additional / airworthiness requirements are applicable to this change. Compliance to CS 23 Amdt. 3 has been shown for all the paragraphs affected by this change, except for the following p (...) | Engineering Order and Certification Program, Doc. No. 2077918-EO, Issue 1, dated 17-Dec-2012-Airplane Flight Manual Supplement, Doc. No. 2077918-AM A, Issue 2, dated 20-Nov-2012, for Garmin GTN 750 and GTN 650-Airplane Flight Manual Supplement, Doc. (...) | This change is limited to the configuration of aircraft and equipments defined in the Engineering Order, Doc. No. 2077918-EO, Section 4- LNAV/VNAV or LPV is not permitted if the holder agrees to permit another person to use this certificate (alter (...) | 20.12.2012 |
| 10043101 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.00764 | | | CESSNA 525 & CESSNA 525A | Installation of a Honeywell KMD 850 System and installation of a TCAS ISky. | | - Flight Manual Supplement 2016515-AM-01, Issue 3, or later EASA Approved Revisions. - Flight Manual Supplement 2016516-AM-01, Issue 3, or later EASA Approved Revisions. | 1. This STC is approved only for the product configurations defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 13.06.2005 |
| 10043321 | SCANDINAVIAN AVIONICS DESIGN GMBH | EASA.A.S.03790 | | | AEROSPATIALE ATR 42-200 AEROSPATIALE ATR 42-300 AEROSPATIALE ATR 42-320 AEROSPATIALE ATR 42-400 AEROSPATIALE ATR 42-500 | Installation of Honeywell KLN-90B System with dual CDr | | See Scandinavian Avionics Design documents: # Certification Documents 2051746 DOC issue 1# Engineering Order 2051746-EO Issue 2# Aeroplane Flight Manual Supplement 2051746-AM Issue 2# Continued Airworthiness 2051746-CA issue 2# Maintenance Manual Supp (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The (...) | 01.07.2009 |
| 10043585 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | DHC-8-401, DHC-8-402 | Dual NAVAERO EFB Class 2 System Provisions | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation must be in accordance with Technical Order Doc. 2068119-TC Issue 3. Operation must be in accordance with AFM Supplement Doc. 2068119-AM A Issue 3 or Doc. 2068119-AM B, Issue 3. Maintenance must be in accordance with MM Supplement Doc. 2068 (...) | This STC does not constitute an Operational Approval. Individual Operational Approvals, including the COTS equipment suitability, must be issued by the competent NAA. Prior to installation of this design change it must be determined that the interrelated (...) | 08.02.2013 |
| 10043766 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | 500 (CITATION) 550 (CITATION I) 550 (CITATION II) 550 (CITATION ULTRA) 550 (CITATION V) 550 (CITATION S/II) | Installation of Universal UNS-1Lw FMS System | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Scandinavian Avionics, AFM Supplement 2080144-AM A, issue 2 dated 23 January 2013; Scandinavian Avionics, Continued Airworthiness 2080144-CA A, Issue 1, dated 13 December 2012; Scandinavian Avionics, Maintenance Manual Supplement 2080144-MM A, issue 1, d (...) | See Scandinavian Avionics AFM Supplement 2080144-AM APrior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce n (...) | 22.02.2013 |
| 10044719 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | 550 CITATION BRAVO 550 CITATION II 550 CITATION S/II | Stand-alone AVIATOR 300 Satcom system installation as identified in Engineering Order 2072808-EO | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Scandinavian Avionics, AFM Supplement 2072808-AM, issue 1 dated 06 July 2012; Scandinavian Avionics, Continued Airworthiness 2072808-CA, Issue 2, dated 09 November 2012; Scandinavian Avionics, Maintenance Manual Supplement 2072808-MM, issue 2, dated 14 S (...) | See Scandinavian Avionics AFM Supplement 2072808-AM | 30.04.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|-------------------|----------------|-------------------------|---|---|---|---|---|---------------|
| 10045018 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | FALCON 2000, 2000EX | THRANE & THRANE AVIATOR 350 SATELLITE COMMUNICATION SYSTEM WITH WIRELESS LOCAL AREA NETWORK. Description as per Associated Technical Documentation | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Documents List in document ref. 2081137 DOC issue 2, dated 24/05/2013, Master Drawings List in document ref. 2081137-DL issue 1, dated 27/03/2013, Certification Plan ref. 2081137-EO issue 2, dated 20/05/2013, Flight Manual Supplement ref. 2081137-A (...) | None | 24.05.2013 |
| 10045534 | SCANDINAVIAN AVIONICS DESIGN GMBH | | REV. 1 | | CESSNA 500 CITATION/CITATION I CESSNA 550 CITATION II CESSNA S550 CITATION S/II | Installation of Garmin Dual GTN-750 System. Revision 1 adds the TAWS limitation to MSN 550-0672 and revokes the initial issue of the approval. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Scandinavian Avionics, AFM Supplement GTN-750, 2081059-AM A, issue 4, dated 12 June 2013-Scandinavian Avionics, AFM Supplement GMA-35, 2081059-AM B, issue 1, dated 25 February 2013-Scandinavian Avionics, AFM Supplement GWX-70, 2081059-AM C, issue 1, (...) | Periodic Structural Inspection: GPS Antenna, Task No. 34-2081059-02 Initial: At next 24 month / 800 hours inspection. Thereafter: at every 24 month / 800 hours inspection. GPS Antenna, Task No. 34-2081059-03 Initial: no later than 11.000 LDG (after instal (...) | 22.08.2013 |
| 10045827 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | ATR 42-200 ATR 42-300 ATR 42-320 ATR 42-400 ATR 42-500 | Installation of Universal UNS FMS System. This STC installs a new fully integrated navigation and flight management system, identified as Universal UNS-1Lw, interfaced with EFIS without Advisory Vertical Guidance displayed on EHSI. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amdt 4 (Cf applicable paragraphs identified through Compliance Record document, ref. 2080925-COR) The requirements for (...) | Master Document list, ref.: 2080925-DOC issue 1, dated 19/07/2013; FMS Configuration with advisory VNAV : Aircraft Flight Manual Supplement, ref.: 2080925-AM A issue 2; FMS Configuration without advisory VNAV : Aircraft Flight Manual Supplement, ref.: 2 (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 19.07.2013 |
| 10045949 | SCANDINAVIAN AVIONICS DESIGN GMBH | | | | AS365 N2 AS365 N3 SA365 N SA365 N1 | RDR-1400C Weather Radar Installation. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 29 Amdt 3 all paragraphs except -, 29.337; FAR 29 Amdt 16, -, 29.561; FAR 29 Amdt 16 The requirements for environmental protection (...) | 2082416-AM issue 3 (RFM SUPPLEMENT) (29.07.2013) or later revisions of the above listed documents approved by EASA, 2082416-DOC issue 1 (Certification documents list) (30.07.2013) 2082416-EO issue 1 (Engineering Order) (30.07.2013) 2082416-COR issue 2 (C (...) | Weather Radar System MUST BE turned to "STBY" or "OFF" when on the ground. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the air (...) | 02.08.2013 |
| 10014565 | SCHERDEL ENGINEERING Ltd. | EASA.A.S.02665 | | | | Change of Empty Mass and Ballast Requirements for the Glider LS1-f | | Doc Number 06-262-001 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "#Associated Technical Documentation#. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 19.02.2007 |
| 10015154 | SCHERDEL ENGINEERING Ltd. | EASA.A.S.03519 | | | DG 400 DG 500 M DG 600M DG 800A SCHEMPP JANUS CM SCHEMPP NIMBUS 4M | Alternate fuel lines for Rotax engine powered glider. | | Scherdel Engineering, Technical Note TN2-Alternate Fuel Lines for Rotax powered Glider, Doc number 08-Alternate Fuel Lines-001, Revision 1 | For all serial numbers. | 11.06.2008 |
| 10015202 | SCHERDEL ENGINEERING Ltd. | EASA.A.S.03599 | | | SCHEMPP CIRRUS | Standard Cirrus: Weight of nonlifting parts increase with winglets. Doc. No. 07-278-005, rev. 2. Doc. No. 07-278-006, rev. 3 | | Flight Manual: pages to be exchanged 2B, 3, 3A, 3B, 11, 12, 1Aa, 12B, 14, 14A, 18A. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 02.10.2008 |
| 10030420 | SCHERDEL ENGINEERING Ltd. | | | | ASK 21, ASK 21 MI | ASK 21 Front Instrument Panel Modification - Modification of the ASK 21 front seat instrument panel to allow more leg space for the pilot | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Doc No 10_ASK21 FIP-001 | None | 15.06.2010 |
| 10038786 | SCHERDEL ENGINEERING Ltd. | | | | 1. DUO DISCUS, 2. DISCUS A 3. DISCUS BT, 4. DUO DISCUS T 5. NIMBUS 4M, 6. VENTUS A 7. STANDARD CIRRUS 8. VENTUS BT, 9. CIRRUS | Cockpit Ventilation Modification | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TN3 Cockpit Ventilation-Additional page to flight manual, section 9a for later revisions of the above listed documents approved by EASA | None | 14.03.2012 |
| 10015763 | SEAPLANES WEST INC. | EASA.IM.A.S.01799 | | TCCA STC SA04-63 | CESSNA 182E CESSNA 182F CESSNA 182G CESSNA 182J CESSNA 182K CESSNA 182L CESSNA 182M CESSNA 182N CESSNA 182P CESSNA 182Q CESSNA 182R | Installation of Aerocet 3400 Amphibious Float Conversion. | | In accordance with Amtech Aeronautical Limited (AAL) Installation Instructions EO333901 dated May 10, 2004, and Master Data List MDL33301 Rev 2 dated November 10, 2004, or later approved revisions. TCCA approved Flight Manual Supplement FMS333902 dat (...) | 1. This STC confers no Operational Approval, that must be sought from the National Airworthiness Authority of the country of registration. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the (...) | 17.07.2006 |
| 10015374 | SECURAPLANE TECHNOLOGIES, Inc. | EASA.IM.A.S.01031 | REV. 1 | TCCA STC SA04-3 ISSUE 3 | ATR 42-300, ATR 42-320, ATR 72-101, ATR 72-102, ATR 72-201, ATR 72-202. | Installation of Hawker sealed Lead-Acid batteries. This STC installs Hawker sealed Lead - Acid batteries P/N 9750W0538 (24V - 40 Ah) and P/N 9750W0350-5 (24 V - 18 Ah). The Revision 1 retains the certified design of the EASA.IM.A.S.01031 and capture (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List, ref. 170-0403-01 Revision L, dated 26 Feb. 2009 Certification Data List, ref.: 170-0410-01 Revision F, dated 29 Oct 2008 Instruction for Continued Airworthiness, ref.: 290-0251-01 Revision D, dated 26 February 2009 or later revisions of (...) | None | 07.01.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|--------------------------|----------------|----------------------|--|---|--|--|--|---------------|
| 10027470 | SECURAPLANE TECHNOLOGIES, Inc. | P- EASA, IM.A.S.03268 | | FAA STC ST01809LA | SEE APPROVED MODEL LIST (AML) | Installation of Hawker 975W0640 Sealed Lead Acid Battery | FAR 25 | Installation in accordance with Approved Master Data List (MDL) Document No 170-0402-01 Revision "-", dated April 13, 2005, or later approved revisions. | This approval is limited to those aircraft listed in the AML and equipped with lead acid battery. | 05.10.2009 |
| 10038222 | SECURAPLANE TECHNOLOGIES, INC. | | | FAA STC SA01696LA | SEE EASA APPROVED MODEL LIST | Installation of Hawker Part Number 9750W0370 or 9750W0370-8Series24Volt 43 Ah Sealed Lead Acid Battery. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation in accordance with Installation Instructions ref 105-1126-01, Rev C as specified by MDL listed in the AML to this STC. Operation in accordance with Flight Manual Supplement as listed in the AML to this STC. Instructions for Continued Airworthiness (...) | None | 30.01.2012 |
| 10044847 | SECURAPLANE TECHNOLOGIES, Inc. | | | FAA STC ST01471LA-D | SEE APPROVED MODEL LIST | Installation of Hawker PN 9750W0370 Series 24Volt/43Amp-hour Sealed Lead-Acid Battery | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC FAA ST01471LA-D, reissued 6th October 2004, with FAA Approved Model List (AML) AML-ST01471LA-D at Revision G-FAA Approved Master Data List (MDL) ref doc. 170-0404-01, Revision E for Cessna Citation Series Aircraft - Certification Data List (...) | None | 10.05.2013 |
| 10013938 | SEFA | EASA, A.S.00352 | REV. 1 | | BEECH B200 | - Installation of calibration workbench ref. STNA "Margouillat1" according to the "02-1104-DIR document". - Installation of calibration workbench ref. STNA "Margouillat1". Revision 1: Extension of STC EASA, A.S.00352 with a minor change to another aircraft (F-GJFE s/n BB1399). | | - Aircraft Flight Manual Supplement ref 02-1104-DIR dated November 2002 or later approved revisions. - 02-1104-DIRREV 1 | 1. Applicable only for F-GJFA s/n BB-1270, F-GJFC s/n BB-1347 and F-GJFE s/n BB1399 aircraft. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previous (...) | 16.02.2009 |
| 10014362 | SEGELFLUGGRUPPE SIEMENS MUENCHEN | EASA, A.S.01629 | | | DIAMOND HK 36 TC | Optional operation with skis instead of wheels in winter One piece for serial-number 36730 | | Appendix to Flight and operating Manual | | 31.07.2006 |
| 10039286 | SEGINUS Inc. | | | FAA STC ST02954CH | BN2A-8/-9/-20/-21/-26/-27 BN2B-20/-21/-26/-27 BN2T/-4R/-4S | Installation of a replacement pitot static probe P/N DU130-24 in accordance with FAA STC ST02954CH. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Drawing EH130-24, Rev. -, dated August 5, 2010 Component Installation Manual SEG130, Rev. -, dated October 1, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subseq (...)) | None | 23.04.2012 |
| 10014058 | SELL GmbH | EASA, A.S.01142 | | | BOEING 767-300 | Replacement of galley G3A P/N10160F23-0000 with a new galley P/N 10160F23A00000 designed by Sell GmbH according to Sell Engineering Order Doc. No. EO WA 151043, Issue May 7 2004 | | Model Drawing 10160F23A00000-Galley Installation Drawing 10160F23A00-Instructions for Continued Airworthiness: Component Maintenance Manual Doc. No. CMM No. 25-01-42 | | 11.02.2005 |
| 10014575 | SELL GmbH | EASA, A.S.02679 | | | BOEING 777-200 BOEING 777-300 | Interior Re-configuration of Forward Galley Complex for Emirates airline. | | Type Investigation Document MPD1023G-25-01 iss. 1 dated 17 January 2007. | Limited to B777-200/300 models serial numbers 29324 and 29325. | 18.01.2007 |
| 10014856 | SELL GmbH | EASA, A.S.03080 | REV. 1 | | BOEING 777-300 | Galley Reconfiguration and Replacement for Emirates Airlines | | Sell Type Investigation Document MPD1023G STC2N issue 1 dated 3 July 2007 | Limited to B777-300 serial numbers: 28680, 28687, 29062, 29063, 29064, 29067, 29395, 29396, 32697, 32699, 32700, 32702 | 31.10.2007 |
| 10046519 | SELL GmbH | | | | A340-600 | A340-600 DLH Galley Relocation. Description Summary : The change is related to the Galley Relocation on LHT A340-600 aircraft for: G23, G24, G25, G26, G27 & Pelmet. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | - Certification Summary : CS-A-2011-066A rev C - Type Definition Document : TDD-1092L-25-06-07-08 rev C - ICA : ICAS-A-2011-066A Rev Newer later revisions of the above listed documents approved by EASA. | 1) Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect on the airworthiness of the product. 2 (...) | 27.09.2013 |
| 10014275 | SERAM SARL | EASA, A.S.01511 | | | CESSNA 337D | Installation of one KX 155A Immunity FM, one GNS430 and G1106A with BRNAV capability | | Flight Manual Supplement Rev 0 dated 01/09/2004 | 1. This STC is limited to Cessna 337D SN 337-1048. 2. Autopilot coupled with GNS 430 forbidden. 3. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility (...) | 23.08.2005 |
| 10042367 | SERAM SARL | | | | SEE EASA APPROVED MODEL LIST | Installation of a Garmin GNS 430/AW/AW system, a GA 35/66 GPS antenna a G1106A CDI indicator, a Bendix/King KX 155 NAV/COM system, a KI-203/4/8/9 VOR/ILS indicator, a KN 62A DME receiver, GPS annunciator lights and AV22, KA60, CI 105 antennas. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/ approval. | AFM supplement FM406R for BRNAV/PRNAV/RNP operations revision 0 dated 30/10/2012; AFM supplement FM406R for IFR operations revision 0 dated 30/10/2012; AFM supplement FM406R for VFR operations, revision 0 dated 11/06/2012; SERAM modification approval (...) | Initial installation of this design change must be performed by authorized SERAM technical personnel. This installation is approved for VFR, IFR, RNAV and RNP. Approach operations without coupling of navigation systems to an autopilot or flight director (...) | 27.11.2012 |
| 10016022 | SHADIN AVIONICS | EASA, IM.A.S.02188 | | ST02115CH | CESSNA 500 CESSNA 550 CESSNA 560 CESSNA 550 | Validation of FAA STC ST02115CH Installation of a dual Shadin "ADC 6000 System RVSM Capable"; hardware provisions for RVSM capability. | | Master Data List DN2807A-002, Rev. N of 30 August 2005 or later approved revision and Report 2807A/AFM Supplements; see "Limitations and Conditions" | This STC does not provide RVSM capability to the aircraft (compliance to RVSM airworthiness requirements are not part of this STC); a separate STC has to be applied. Airplane Flight Manual Supplement, Shadin Document No. DN2807A-006, Revision B, dated (...) | 07.12.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------------|-------------------|----------------|---------------------------|---|---|--|---|--|---------------|
| 10016036 | SHADIN AVIONICS | EASA.IM.A.S.02205 | | FAA STC SA02098CH | CESSNA 501 | Validation of FAA STC SA02098CH # #Installation of dual Shadin ADC 6000 Air Data Computer System# for RVSM capability | | Shadin Master Data List Document No. DN2807B-002, Rev. K or later FAA approved Revision FAA approved Shadin AFM-S Doc. No. DN2807B-006, Rev. A, dated 01.09.2005 or later FAA approved Revision for aircraft modified with ADC-6000R, P-No 962861-00- 00 or (...) | The installer must determine, if this design is compatible with previously approved modifications. If the holder agrees to permit another person to use this certification or alter a product, the holder must give the other person written evidence of that permission. | 10.04.2007 |
| 10034421 | SHADIN AVIONICS | | | FAA STC SE578GL | SEE EASA APPROVED MODEL LIST | Fuel Flow Transducer Installation / Installation of Shadin fuel flow transducer in the fuel line between the fuel control unit (FCU) and the fuel injector. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC SE578GL or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None. | 04.04.2011 |
| 10034457 | SHADIN AVIONICS | | | FAA STC SH582GL | BELL 206B, B-1, L, L-1, L-3 BELL 407 | Installation of Digital Fuel Flow Meter. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 4045 revision E dated 15 December 2010 - Installation Instructions for Bell 206 Models; 4119 revision C dated 15 December 2010 - Installation Instructions for Bell 407. | None. | 08.04.2011 |
| 10015809 | SHOWA AIRCRAFT USA INC | EASA.IM.A.S.01878 | | ST01621SE | BOEING 747-400 | Installation of an Upper Deck Galley on Boeing | | Data package containing the following documents, and additional supporting data: 1. Drawing List (60305-DL04 Rev A)2. 8110-3 FAA Approvals GAP Galley Inst. B747-4003. FAA concurrence letter for DF110 oven4. Elec. Load Summary Report (6030 (...)) | | 18.12.2006 |
| 10015846 | SIDDHIS AERONAUTICAL DESIGN CORP. | EASA.IM.A.S.01946 | REV 3 | TCCA STC SA97-71, ISSUE 6 | BOMBARDIER CL-600-1A11 (CL- 600 BOMBARDIER CL-600-2A12 (CL- 601 BOMBARDIER CL-600-2B16 (601- 3A BOMBARDIER CL-600-2B16 (601- 3R BOMBARDIER CL-600-2B16 (604) BOMBARDIER CL-600-2B19 (SERIES | Installation of Fin Mounted Radome/ Antenna System (Structural Provisions only). | | 2. Transport Canada STC SA97-71 Issue 6 dated March 28, 2007 or later approved revision. 3. SIDDHIS Aeronautical Design Modification Summary MS05272 Revision 3 or later approved revision. | 1. The installation shall be visually inspected for external damage every 1200 hours and for lightning damage every flight in severe weather. 2. The total weight of the antenna and the installed equipment shall not exceed 18.1 kg. 3. The additional height clearance of 0.17 meters is required 4 | 29.05.2007 |
| 10015842 | SIERRA INDUSTRIES Ltd. | EASA.IM.A.S.01937 | | FAA STC ST09559AC | CESSNA C501 | Installation of Williams/ Rolls FJ44-2 Engines in accordance with FAA STC ST09559AC and Sierra Ind. Drawing SI360-000, dated 27.02.2002. or later FAA approved Revision. | | Sierra Ind. Drawing SI360-000, dated 27.02.2002, or later FAA approved Revision. FAA approved Sierra Ind. AFM-S Doc. No. SRW137, dated 14.03.2002 or later FAA approved Revision. | The installer must determine, if this design is compatible with previously approved modifications. If the holder agrees to permit another person to use this certification to alter a product, the holder must give the other person written evidence of that permission. | 08.03.2007 |
| 10015851 | SIERRA INDUSTRIES Ltd. | EASA.IM.A.S.01951 | | ST09630AC | CESSNA 501 | Installation of Meggitt Engine Instrument Display System according to FAA STC ST09630AC. | | Sierra Ind. Drawing SI400-001, Rev. A, dated 17.03.2006, or later FAA approved Rev. FAA approved Sierra Ind. AFM-S Doc. No. 115- 601, Rev. 1, dated 15.10.2004 or later FAA approved Revision for Cessna 501 equipped with Williams FJ44 engines FAA approved (...) | The installer must determine, if this design is compatible with previously approved modifications. If the holder agrees to permit another person to use this certification to alter a product, the holder must give the other person written evidence of that permission. | 15.03.2007 |
| 10016178 | SIERRA INDUSTRIES Ltd. | EASA.IM.A.S.02407 | | FAA STC SA29NW | CESSNA 401 CESSNA 402 | Robertson STOL Modification/Installation of fowler flaps, flap actuated droop ailerons, flap actuated trim and anti-servo rudder trim tab in accordance with Drawing List No. 24 | | As referred in FAA STC SA29NW FAA approved Robertson AFM-S, or later FAA approved Revisions Robertson Drawing List No. 24 | As referred in FAA STC SA29NW/Max. Flap Extension Speed VFE: 10° Flap 195 mph (CAS) 30° Flap 145 mph (CAS) Min. Single Engine Control Speed VMC : 10° Flap 83 mph (CAS) Flap up (| 19.03.2008 |
| 10016339 | SIERRA INDUSTRIES Ltd. | EASA.IM.A.S.02649 | | FAA STC SA80RM | CESSNA 550 | Installation of Extended Range Fuel Tank and relocation of Air Pressure Bulkhead. | | MDL 91800 # Master Drawing and Data List, Document FMS-912 # Airplane Flight Manual Supplement for Citation 550 Document MMS-911 # Maintenance Manual Supplement for Citation 550 Document WB-912 # Weight and Balance Data for Citation 550 | Cessna Service Bulletin 34-15 or 34-23 (ref. Note 7 in TCDSA22CE, latest rev.) must be incorporated unless modified aircraft with full cabin tank fuel, minimum crew and required ballast is within maximum zero fuel weight and C.G. limits. | 11.06.2008 |
| 10016340 | SIERRA INDUSTRIES Ltd. | EASA.IM.A.S.02650 | | FAA STC SA174RM | CESSNA 550 | Installation of an extended width cabin door. | | Master Doc. List MDL 91500, Report 151-1 # Cessna Citation Extended Door Structural Analysis#, Rev. A dated 2-25-88 | | 03.07.2008 |
| 10016443 | SIERRA INDUSTRIES Ltd. | EASA.IM.A.S.02788 | | FAA STC SA1689WE | CESSNA 172P | Installation of drooped ailerons in flaps down mode, contoured wing leading edges, except for 172M, 172N, and 172P), and wing stall fences. | | In accordance with FAA sealed Robertson Aircraft Drawing List No 10. | 1. Floatplane Models 172A thru 172H are only eligible with Franklin Engines 6A-350-C1 per STC SA340NW (215HP) or 6A-350-C2 per STC SA1334WE (180 HP). 2. Fixed tabs as shown on R/STOL Systems, Inc. Drawing 10-100 required on Floatplane Model 172E (...) | 13.05.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|--|--|--|---|---|---------------|
| 10016447 | SIERRA INDUSTRIES Ltd. | EASA.IM.A.S.02793 | REV. 1 | FAA STC SA8829SW | CESSNA 500 CESSNA 550 CITATION II | Installation of Inertial Reel Assy.'s on Pilot and Co-Pilot/ShoulderHarnessIn accordance with Sierra Master Drawing List, SI220-000, rev. A, dated 28 March 1994./Validation of FAA approval STC SA8829SW. This is an amendment to EASA.IM.A.S.02793 to include Cessna Model 500 as an approved model. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Sierra Master Drawing List, SI220-000, rev. A, date 28 March 1994; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None. | 06.12.2012 |
| 10016463 | SIERRA INDUSTRIES Ltd. | EASA.IM.A.S.02809 | | FAA STC SA8437SW | CESSNA 550 CITATION II CESSNA S550 (CITATION II) | Installation of radome quick release mechanism.FAA STC SA8437SW. | | Sierra Industries Inc. Drawing List No. SI160-000, Rev. B, dated 13 July 1992 or later approved revisions of this document approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision). This STC extends the (...) | | 28.03.2008 |
| 10016464 | SIERRA INDUSTRIES Ltd. | EASA.IM.A.S.02810 | REV. 1 | FAA STC ST09011SC | CESSNA 500 CESSNA CITATION 550 AND S550 | Pilot's Control Wheel Mounted Alternate Gear Horn Silence Switch.In accordance with Sierra Master Drawing List, SI260-000, rev. IR, date 25 Jan 1994. Validation of FAA approval STC ST09011SC. Amend EASA.IM.A.S.02810 to include the Cessna Model 500 as an approved model. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Ref. LBA STC nr. STA 1309 of 30 January 2003; Sierra Master Drawing List, SI260-000, rev. IR, date 25 Jan 1994; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 27.02.2013 |
| 10016465 | SIERRA INDUSTRIES Ltd. | EASA.IM.A.S.02811 | REV. 1 | FAA STC SA09229SC | CESSNA 500 CESSNA 550 CITATION II CESSNA S550 (CITATION S/II) | Installation of Gas Cylinder Nose baggage Door Openers in accordance with Sierra Master Drawing List, SI380-000, rev. C, date 24 Sept 1996. Validation of FAA approval STC SA09229SC. This is an amendment to EASA.IM.A.S.02811 to include Cessna Model 500 as an approved model. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Sierra Master Drawing List, SI380-000, rev. C, date 24 Sept. 1996; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None. | 06.12.2012 |
| 10030649 | SIERRA INDUSTRIES Ltd. | | REV. 1 | FAA STC SA3634SW | 206H, T206H LANDPLANES P206 A, D, E LANDPLANES U206 LANDPLANES A, B, D U206 LANDPLANES E, F, G | Auxiliary Fuel System: Installation of three auxiliary bladder type fuel tanks in each wing in accordance with FAA STC SA3634SW, revision 4, dated 30.03.2010. This EASA STC is a revision of LBA STC (EMZ) 0393/673, 673A, dated 05.05.1993. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Drawing List No. SI10600, Revision N, dated January 27, 2006- Installation Instruction No. SI10665, Revision D, dated July 18, 1986- Airplane Flight Manual Supplements for Cessna P206 A, D and E model landplanes (S/N P206-0161 through P206-00647); (...) | Cessna P206 A, D and E model landplanes (S/N P206-0161 through P206-00647); Cessna U206 model landplanes A, B, D (-> ->) E, F, G (S/N U206-0438 through U206-04649); Cessna 206H and T206H model landplanes (S/N 206-08001 to present and S/N T206-08001 to (...) | 17.10.2012 |
| 10036302 | SIERRA INDUSTRIES Ltd. | | | FAA STC ST02363LA | HAWKER 800XP | Installation of Honeywell TFE731-50R Engines. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CAR 4b and FAR 25 as defined on the FAA STC certificate ST02363LA. This certificate/ app (...) | Installation in accordance with Sierra Industries, Ltd Master Drawing List No 201-005, Rev F dated July 13, 2011. Operation in accordance with Sierra Industries, Ltd Aircraft Flight Manual Supplement No 201-191 (-> ->) dated August 09, 2011. Instructions (...) | Limited to Hawker 800XP aircraft fitted with Honeywell SPZ-2000 Avionics System, and Winglets installed under Hawker Beechcraft (HBC) Mod Kit 140-1703 -Service Bulletin 57-3810, EASA approval ref EASA.IM.A.C.01451. | 30.08.2011 |
| 10037473 | SIERRA INDUSTRIES Ltd. | | | FAA STC SA1435WE | 180H | Installation of drooped ailerons in flap down mode, re-contoured wing leading edge in accordance with FAA STC SA1435WE. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Owner's Manual Supplement OMS 10-4 Drawing List No. 10, dated June 25, 1985 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). Maintenance Instructions SMM 10-1 | S/N 18051446 through 18052284 only. Installation of SC-100 Stall Warning system is not included. CG range: straight line between points- Amphibian with (+4.2) to (+43.9) @ 2950 lb. - EDO 597-2790 (+36.0) to (+43.9) @ 2258 lb. or less. Control Surface Mo (...) | 29.11.2011 |
| 10042597 | SIERRA INDUSTRIES Ltd. | | | FAA STC SA09630AC | CESSNA 500 | Installation of Meggitt Engine Indicating Display System. In accordance with Sierra Master Drawing List, SI400-001, rev. B, date 29 Jan 2008. Validation of FAA approval STC SA09630AC. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Sierra Master Drawing List, SI400-001, rev. B, date 29 Jan 2008; Sierra AFM Supplement, ref 115-601, rev 2, date 7 May 2007; Sierra Maintenance Manual Supplement, ref SM-400 MMS, issue 1, date 12 May 2004; Sierra Instruction for Continued Airworthiness, (...) | None. | 10.12.2012 |
| 10016389 | SIERRACIN/SYLMAR | EASA.IM.A.S.02718 | | FAA STC ST01840NY | BOMBARDIER DHC-8-100 SERIES BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-300 SERIES | Installation of Sierracin/Sylmar Windshield Assembly P/N 802600-11/-12, Ref.: FAA STC ST01840NY. | | Installation must be performed in accordance with Aero Consulting Services Ltd. Modification Data Summary D641000, Revision N/C, dated 4 November 2003. Instructions for Continued Airworthiness/ Maintenance to be performed in accordance with Component Maintenance Manual 56-10 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 14.08.2008 |
| 10016390 | SIERRACIN/SYLMAR | EASA.IM.A.S.02719 | REV. 1 | FAA STC ST01839NY | BOMBARDIER DHC-8-100 SERIES BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-300 SERIES BOMBARDIER DHC-8-400 SERIES | Installation of Sierracin/Sylmar Windshield Assembly P/N 802600-09/-10, Ref. FAA STC ST01839NY | | Installation must be performed in accordance with Aero Consulting Services Ltd. Modification Data Summary D626000, Revision N/C, dated 27 October 2003. Instructions for Continued Airworthiness/ Maintenance to be performed in accordance with Component Maintenance Manual 56-10-1 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 15.08.2008 |
| 10016716 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01010 | | FAA STC SR00836NY | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of Cargo Compartment Baggage Liner. | | -FAA STC SR00836NY -Keystone Helicopter Cor. Installation Drawing IDL00081, Revision A, dated January 29, 1998. | 1. Keystone Helicopter Corp. Maintenance Manual S-76A, B&C Baggage Liner, Report No MM00081 dated February 17, 1997, as required. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the para (...) | 11.04.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|--|---|---|--|--|---------------|
| 10016717 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01011 | | FAA STC SR01131NY-D | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of Luminescent Systems Helicopter Emergency Egress System (HEELS) | | -FAA STC SR01131NY-D.-Keytech Installation Drawing List IDL00565 Revision A dated April 12 2002 or later approved revision.-RFM Supplement No FM00565, no revision, approved December 10 2002. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 19.04.2005 |
| 10016718 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01012 | | FAA STC SR01593NY-D | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of Keystone Helicopter Utility Interior Panels | | -FAA STC No SR01593NY-D.-Keytech Installation Drawing List (DL00569 Revision B dated July 13 2004 or later approved revision). | 1. Keystone Helicopter Instructions for Continued Airworthiness Report No IC00735 Rev. dated June 11, 2004 or later approved revision.2. It is a prerequisite to have previously approved brackets installed for each individual panel installation as den (...) | 22.04.2005 |
| 10016733 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01057 | | FAA STC SR00033NY | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of Grease Fittings. | | -FAA STC n° SR00033NY.-Keytech Corporation Installation Drawing List Report No IDL93-32 revision B dated January 18, 1999 or later approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.04.2006 |
| 10016734 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01058 | | SR01596NY-D | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Universal Avionics Systems Corporation UNS-1 Multi-Mission Management Systems (MMS) with SCN 901.X Software | | Keytech Installation drawing list No. IDL00668 Rev D Dated October 17 2003 or later approved revision.Keystone Helicopter Corp. RFM Supplement No. FM00668 Revision 1 dated February 2006 or later approved revision. | | 06.03.2006 |
| 10016735 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01059 | | FAA STC SR01132NY-D | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Honeywell KT870 Traffic Advisory (TAS) displays on a Honeywell KD550 Multifunctional Display (MFD)Ass.-Report ER00566Certification PlanRFM Supplement FM00566 | | Report ER00566 Certification PlanRFM Supplement FM00566 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.11.2005 |
| 10016736 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01060 | | FAA STC SR01608NY-D | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of Aircell ST 3100 Satellite Telephone System | | Keystone Helicopter installation drawing list IDL00783, Revision A dated 5/18/04 or later approved revisionKeystone Helicopter AFM Supplement No FM00783 Rev 1 dated 24 August 2005or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 23.02.2006 |
| 10016753 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01082 | | SR00688NY | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of Medical Interior | | Keystone Helicopter Installation Drawing List IDL00136 Revision C dated March 15, 1999 or later approved revisionKeystone Helicopter RFM Supplement FM00136, FM00136.2 & FM00136.4 dated September 26, 1997 or later approved revision for Sikorsky S-76 A (...) | | 12.05.2006 |
| 10016773 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01108 | | FAA STC SR01958NY-D | SIKORSKY S-76C | Installation of Rockwell Collins TDR-94D Transponder with Elementary Surveillance | | Installation Drawing List IDL00926 Rev A dated May 11, 2005or later approved revisionFlight Manual Supplement FM00926 rev - dated May 20, 2005 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 13.03.2006 |
| 10016774 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01109 | | FAA STC SR01960NY-D | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of baggage access panel (ground use only), forward facing seat, seat pan and wet floor. | | Installation Drawing List IDL00921 rev " " dated May 16, 2005 or later approved revisionFlight Manual Supplement FM00921 rev - dated May 24, 2005 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 13.03.2006 |
| 10016821 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01186 | | SR01134NY-D | SIKORSKY S-76C | Installation of Bristol Helicopters Ltd Automatic floatingDeployment System (AFDS) in accordance with Keytech Installation drawing list IDL00601 Rev. dated 11/27/02 | | Keystone helicopter Corp. FAA approved rotorcraft flight manual Supplement No. FM00601, no revision, FAA approved December 10, 2002, is required with this modification | This STC is valid only for S76C Sikorsky helicopter model from S/N 760511 and subsequentThis STC should not be incorporated in any aircraft of this specific model on which other approved modifications are incorporated, unless is determined that the (...) | 29.03.2006 |
| 10016855 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01231 | | FAA STC SR01140NY-D | SIKORSKY S-76C | Installation of a Cyclic Control Stick Guard | | Keytech Installation Drawing List IDL00644 Revision - datedApril 17 2003 or later approved revision.Keystone Helicopter AFM Supplement 00644 no Revision dated April 25 2003 or later approved revision. | | 21.11.2006 |
| 10016863 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01239 | | FAA STC SR01135NY-D | SIKORSKY S-76C | Installation of a Chadwick-Helmuth Vibration Monitoring System | | Keystone Helicopter Installation Drawing List IDL00571 Revision -, dated January 13, 2003 or later approved revision | | 26.09.2006 |
| 10016864 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01240 | | FAA STC SR02110NY-D | SIKORSKY S-76A SIKORSKY S-76C | Installation of a Honeywell Chadwick Model 15175 VXP System | | Keystone Helicopter Installation Drawing List IDL00892 Revision A dated November 21, 2005 or later approved revision | | 26.09.2006 |
| 10016879 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01256 | REV. 1 | FAA STC SR02408NY-D | S-92A | Installation of Spectrolab Nightsun II XP SearchlightNote: This certificate revises and supersedes STC no. EASA.IM.R.S.01256, aiming to update the Associated Technical Documentation as listed below | The Certification Basis for the original product remains applicable to this certificate/ approval | FAA approved Rotorcraft Flight Manual Supplement No.FM01231 rev. 2dated September 16, 2008or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)- Ke (...) | None | 22.11.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|----------------|---|---------------------|--|------------|---------------|
| 10016880 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01257 | | FAA STC SR02411NY-D | SIKORSKY S-92A | WESCAM MX-15i FLIR Installation on a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document Reference DL01232, rev. B or later approved revisions. Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01232, rev. - or later approved revisions. Instructions for (...) | | 02.05.2007 |
| 10016881 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01258 | | FAA STC SR02412NY-D | SIKORSKY S-92A | Saab airborne Transponder System Installation in a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document Reference DL01233, rev. B or later approved revisions. Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01233, rev. - or later approved revisions. Instructions for (...) | | 02.05.2007 |
| 10016882 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01259 | | FAA STC SR02403NY-D | SIKORSKY S-92A | Chelton Tetra Transceiver Provision Installation in a Sikorsky S92 Helicopter. | | Installation in accordance with Keystone Document Reference DL01234, rev. B or later approved revisions. Instructions for Continued Airworthiness in accordance with Keystone Document Reference IC01234, rev. - or later approved revisions. | | 02.05.2007 |
| 10016883 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01260 | | FAA STC SR02400NY-D | SIKORSKY S-92A | Helicopter Emergency Egress Lighting System (HEELS) Installation in a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document Reference DL01236, rev. - or later approved revisions. Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01236, rev. - or later approved revisions. Instructions for (...) | | 02.05.2007 |
| 10016884 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01261 | | FAA STC SR02404NY-D | SIKORSKY S-92A | Scene Lights Installation in a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document Reference DL01237, rev. - or later approved revisions. Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01237, rev. - or later approved revisions. Instructions for (...) | | 02.05.2007 |
| 10016885 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01262 | | FAA STC SR02133NY-D | SIKORSKY S-92A | Side Facing Utility Seat Installation in a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document Reference DL01238, rev. B or later approved revisions. Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01238, rev. - or later approved revisions. Instructions for (...) | | 02.05.2007 |
| 10016886 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01263 | | FAA STC SR02414NY-D | SIKORSKY S-92A | GKN Sliding Entry Door Installation in a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document Reference DL01270, rev. A or later approved revisions. Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01270, rev. - or later approved revisions. Instructions for C (...) | | 02.05.2007 |
| 10016887 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01264 | | FAA STC SR02405NY-D | SIKORSKY S-92A | Logo Light Installation in a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document Reference DL01272, rev. - or later approved revisions. Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01272, rev. - or later approved revisions. Instructions for (...) | | 02.05.2007 |
| 10016888 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01265 | | FAA STC SR02407NY-D | SIKORSKY S-92A | Auxiliary DC Power System Installation in a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document Reference DL01299, rev. - or later approved revisions. Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM-1299, rev. - or later approved revisions. Instructions for (...) | | 02.05.2007 |
| 10016889 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01266 | | FAA STC SR02121NY-D | SIKORSKY S-92A | Shelf Installation in Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document Reference DL01334, rev. - or later approved revisions. Instructions for Continued Airworthiness in accordance with Keystone Document Reference IC01334, rev. - or later approved revisions. | | 02.05.2007 |
| 10016890 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01267 | | FAA STC SR02132NY-D | SIKORSKY S-92A | Ramp Gap Bridge Installation in a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document Reference DL01342, rev. - or later approved revisions. Instructions for Continued Airworthiness in accordance with Keystone Document Reference IC01342, rev. - or later approved revisions. | | 02.05.2007 |
| 10016891 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01268 | | FAA STC SR02124NY-D | SIKORSKY S-92A | Jettisonable Cabin Window Installation on a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document Reference DL01229, rev. D or later approved revisions. Instructions for Continued Airworthiness in accordance with Keystone Document Reference IC01229, rev. A or later approved revisions. | | 02.05.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|----------------------|--|---|---------------------|--|---|---------------|
| 10016892 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01269 | | FAA STC SR02401NY-D | SIKORSKY S-92A | SAR Operators Seat Installation on a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document ReferenceDL01230, rev. A or later approved revisions.Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01230, rev. - or later approved revisions.Instructions for (...) | | 02.05.2007 |
| 10016893 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01270 | | FAA STC SR02413NY-D | SIKORSKY S-92A | Wulfsberg Flexcomm II Communications Management System Installation on a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document ReferenceDL01221, rev. A or later approved revisions.Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01221, rev. - or later approved revisions.Instructions for (...) | | 02.05.2007 |
| 10016894 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01271 | REV. 1 | FAA STC SR02433NY-D | SIKORSKY S-92A | NAT Loudhailer System Installation on a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document ReferenceDL01222, rev. A or later approved revisions.Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01222, rev. - or later approved revisions.Instructions for (...) | | 29.10.2007 |
| 10016895 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01272 | | FAA STC SR02406NY-D | SIKORSKY S-92A | Honeywell Primus 701A Weather Radar Installation on a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document ReferenceDL01224, rev. A or later approved revisions.Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01224, rev. - or later approved revisions.Instructions for (...) | | 02.05.2007 |
| 10016896 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01273 | | FAA STC SR02410NY-D | SIKORSKY S-92A | SAR Audio System Installation on a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document ReferenceDL01223, rev. C or later approved revisions.Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01223, rev. - or later approved revisions.Instructions for (...) | | 02.05.2007 |
| 10016897 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01274 | | FAA STC SR02126NY-D | SIKORSKY S-92A | Skytrac System ISAT-100 Installation on a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document ReferenceDL01225, rev. B or later approved revisions.Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01225, rev. - or later approved revisions.Instructions for (...) | | 02.05.2007 |
| 10016898 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01275 | | FAA STC SR02409NY-D | SIKORSKY S-92A | Polycorn Wireless Intercom System Installation on a SikorskyS-92 Helicopter. | | Installation in accordance with Keystone Document ReferenceDL01226, rev. - or later approved revisions.Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01226, rev. - or later approved revisions.Instructions for (...) | | 02.05.2007 |
| 10016899 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01276 | | FAA STC SR02415NY-D | SIKORSKY S-92A | Chelton 935-2 VHF Direction Finder Installation on a Sikorsky S-92 Helicopter. | | Installation in accordance with Keystone Document ReferenceDL01227, rev. - or later approved revisions.Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01227, rev. - or later approved revisions.Instructions for (...) | | 02.05.2007 |
| 10016900 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01277 | | FAA STC SR02402NY-D | SIKORSKY S-92A | Search and Rescue (SAR) Interior Installation on a SikorskyS-92 Helicopter. | | Installation in accordance with Keystone Document ReferenceDL01228, rev. B or later approved revisions.Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01228, rev. - or later approved revisions.Instructions for (...) | | 02.05.2007 |
| 10016910 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01291 | | FAA STC SR02129NY-D | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of Keystone Helicopter Corporation #Silencer# Wall panel System. Classified as Basic STC by FAA. | | IDL00916 rev # dated 15 February 2007 # Installation Drawing ListIC00916 rev # dated 18 April 2006 # Instructions for Continued Airworthiness | Compliance with FAR 29.812 has not been established for theinstallation of this system. An approved emergency lightingsystem is required.Compliance with FAR 29.783(c)(e), 29.803(a)(b), 29.807(a)(b), 29.809(c), 29.811, 29.813(a), 29.1301(a)(d) and 29 (...) | 15.03.2007 |
| 10016913 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01294 | | FAA STC SR02131NY-D | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of Keystone Helicopter Corporation Deluxe VIP Interior. | | # IDL01217 rev # Installation Drawing List # dated 17 February 2007 or later approved revision# IC01217 rev # Instructions for Continued Airworthiness # dated 8 May 2006 or later approved revision# FM01217 rev # Flight Manual Supplement # dated 18 February 2007 or later approved revision | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 26.07.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|------------------------------|----------------|----------------------|--|--|--|--|--|---------------|
| 10016914 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01295 | | FAA STC SR02130NY-D | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of Garmin GMX 200 Multi Function Display. This STC has been classified as "BASIC" by the FAA and has been approved under the PTVP. The flight manual has been reviewed and is acceptable. | | IDL01357 rev # dated 11 February 2007 # Installation Drawing List or later approved revision; IC01357 rev # dated 21 August 2006 # Instructions for Continued Airworthiness or later approved revision; FM01357 rev 2 dated 18 June 2007 # Flight Manual Supplement or later approved revision | | 03.07.2007 |
| 10016915 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01296 | | FAA STC SR02123NY-D | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of Provisions for a EuroAvionics EuroNav IV System. STC SR02123NY-D classified as Basic by the FAA through FAA Order 8110.52. | | IDL01318 rev B dated 22 December 2006 # Installation Drawing List; IC01318 rev # dated 22 August 2006 # Instructions for Continued Airworthiness | The FAA has not certified, and EASA has not validated, the equipment for which these provisions are intended. Additional approvals are required to install this equipment. The equipment installation must be evaluated to ensure that it complies with the applicable Certification Specifications. | 19.03.2007 |
| 10016916 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01297 | | FAA STC SR02122NY-D | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of High Intensity Strobe Lights. STC Classified as Basic by FAA. | | IDL01319 rev B dated 15 December 2006 # Installation Drawing List; IC01319 rev # dated 27 July 2006 # Instructions for Continued Airworthiness; FM01319 rev # dated 21 December 2006 # Flight Manual Supplement | | 15.03.2007 |
| 10016946 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01340 | | FAA STC SR01976NY-D | SIKORSKY S-92A | Installation of Keystone Helicopter Auxiliary Gauges and Relocation of Fill Ports for the Low Float Pressure Warning System. | | # Installation in accordance with Keystone Document Reference IDL00972 rev C dated 8/29/2005, or later approved revisions. # Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM00972, rev. # dated 12-19-2005, o (...) | | 29.06.2007 |
| 10016981 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01384 | | FAA STC SR02417NY-D | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of Honeywell KHF 1050 HF Communication System. | | Installation Drawing List IDL01407. | Refer to FAA STC SR02417NY-D. | 04.12.2007 |
| 10016982 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01386 REVISION 1 | REV. 1 | FAA STC SR02152NY-D | SIKORSKY S-92A | Installation of a Cockpit Pressure Refuelling Control in a Sikorsky S-92A. | | IDL01367 Drawing List. Revision B dated 9 January 2007 or later approved revision; FM01367 Flight Manual Supplement Revision dated 9 January 2007 or later approved revision; IC01367 Instructions for Continued Airworthiness dated 8 December 2006 or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The inst (...) | 09.11.2007 |
| 10016983 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01387 | | FAA STC SR02426NY-D | SIKORSKY S-76A SIKORSKY S-76B SIKORSKY S-76C | Installation of a Replacement Concorde Lead Acid Battery. | | Installation Drawing List IDL01379 Rev. RFM FM01379. | | 05.02.2008 |
| 10016997 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01406 | REV. 1 | FAA STC02436NY-D | SIKORSKY S-92A | Installation of a High Intensity Strobe Lights. This STC has been classified as BASIC by the FAA and has been approved under the PTVP. | | FM01368 revision dated 24 January 2008 or later approved revision Flight manual Supplement; IC01368 revision dated 20 November 2006 or later approved revision Instructions for Continued Airworthiness; IDL 01368 revision A dated 11 January 2008 or I (...) | | 19.03.2008 |
| 10017018 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01436 | REV. 1 | FAA STC SR01949NY-D | EUROCOPTER FRANCE AS 365N1 EUROCOPTER FRANCE AS 365N2 EUROCOPTER FRANCE AS 365N3 | Installation of a Honeywell Chadwick Model 15175 VXP System. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | 1. FM00771 revision - dated 1/11/2005 or later approved revision - Flight manual Supplement. 2. IC00771 revision - dated 11/09/2004 or later approved revision - Instructions for Continued Airworthiness. 3. IDL 00771 revision - dated 11/09/2004 o (...) | none | 29.06.2011 |
| 10017040 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01464 | | FAA STC SR02435NY-D | SIKORSKY S-92A | Installation of a Rockwell Collins SAR-126 Direction Finder System. This STC has been classified as #BASIC# by the FAA and has been approved under the PTVP. | | FM01366 revision # dated 4 January 2008 or later approved revision # Flight manual Supplement; IC01366 revision B dated 30 September 2007 or later approved revision # Instructions for Continued Airworthiness; IDL 01366 revision C dated 26 November 200 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 10.07.2008 |
| 10017070 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01506 | | FAA STC SR02577NY-D | SIKORSKY S-76C | Keystone Helicopters STC SR02577NY-D: Introduction of a cockpit digital video recorder system to meet JAR-OPS 3 flight data recorder requirements on applicable Sikorsky S-76C series aircraft. | | 1. Keystone Rotorcraft Flight Manual Supplement FM01394 Rev.1"; Cert. Plan ER01394 Rev.A"; Drawing Lists IDL 01394 Rev.A & RDL 01394 Rev.-"; Instructions for Cont.A/w; IC01394 Rev.A" or later approved revisions of the above documents | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 19.02.2009 |
| 10017076 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01519 | | FAA STC SR02573NY-D | SIKORSKY S-92A | Installation of Main Rotor Blade Recognition Lights. | | - IDL01701 revision # dated 9 October 2008 or later EASA approved revisions # Installation Drawing List. - IC01701 revision # dated 14 March 2008 or later EASA approved revision # Instructions for continued Airworthiness; - FM01701 revision # d (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 02.12.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|---------------------|----------------|-------------------------------|--|---|--|---|--|---------------|
| 10017083 | SIKORSKY GLOBAL HELICOPTERS Inc. | EASA.IM.R.S.01550 | | FAA STC SR02208NY-D | SIKORSKY S-92A | Installation of lavatory. | | LDL00845 revision A dated 19 December 2005 or later approved revisions # Installation Drawing List # Configuration 1; LDL00845.1 revision A dated 19 May 2006 or later approved revisions # Installation Drawing List # Configuration 2; IC00 (...) | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 30.04.2009 |
| 10029325 | SILVERHAWK CONVERSIONS | P-EASA.IM.A.S.02962 | | FAA STC SA10741SC | BEECHCRAFT KING AIR C90, E90, C90A AIRCRAFT | Installation of Pratt and Whitney Canada PT6A - 135A Engines. | CS-23.B NSC | Installation of Pratt & Whitney PT6A-135 engines in accordance with Drawing No. 70101-00, Rev. IR, dated 06. June 2007 or later approved revision. Operation must be in accordance with Aircraft Flight Manual Supplement No. AFMS 70101-C90 dated 20. Septem (...) | None | 18.03.2010 |
| 10015331 | SIMMONDS PRECISION PRODUCTS, INC | EASA.IM.A.S.00345 | | FAA STC ST00089BO | BOEING 727-200 SERIES | Installation of either the Goodrich Corporation Fire Detection and Suppression system, or the Fire Detection only system in accordance with Goodrich Corporation Master Drawing List T3008-MDL-0101, Revision A, dated October 31, 2003 or later FAA-appr (...) | | | 1. The suppression feature of Fire Detection and Suppression system has been certified for a maximum of 60 minutes from the activation of the suppression agent. Aircraft diversion time, including landing, should not exceed 60 minutes. 2. This (...) | 30.11.2004 |
| 10015770 | SIMMONDS PRECISION PRODUCTS, INC | EASA.IM.A.S.01809 | | FAA STC ST00043BO | BOEING 737-100 BOEING 737-200 BOEING 737-300 BOEING 737-400 BOEING 737-500 | Installation of a Simmonds Precision 50203 series Fuel Quantity Harness in accordance with Simmonds Precision Installation Document NO E-3203, dated May 24, 1995, or later approved revision | | Simmonds Precision Installation Document NO E-3203, dated May 24, 1995, or later approved revision | Compatibility of this design change with other previously approved modifications must be determined by the installer. | 22.01.2007 |
| 10016158 | SIMMONDS PRECISION PRODUCTS, INC | EASA.IM.A.S.02377 | | FAA STC ST00235BO | BOEING 737-400 | Installation of the Goodrich Corporation Fire Detection (Class #E#) | | Goodrich Corporation Master Drawing List T3078-MDL-2001, Revision H, dated May 30, 2006 or later approved revision | 1. The installation of this equipment is only applicable on 737-400 aircraft modified to full freighter configuration per Pemco STC Numbers SA2968SO, SA2969SO, SA2970SO and SA2971SO inclusively. 2. Goodrich Corporation Airplane Flight Manual Supplement (...) | 07.05.2007 |
| 10016338 | SIMMONDS PRECISION PRODUCTS, INC | EASA.IM.A.S.02647 | | FAA STC ST00092BO | AIRBUS A300 | In-Tank Fuel Quantity Indicating System (FAA STC ST00092BO validation). This STC installs, within the aircraft fuel tanks, a new FQIS sensors and the associated wire harnesses. | | Master Drawing List, ref : T3012-MDL-0101 Revision D, dated 20 March 2007. (or any later EASA approved revision) Instruction for Continued Airworthiness, ref : T3012-0005-0101 Revision A, dated 07 February 2007. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 02.10.2008 |
| 10016713 | SIMPLEX MANUFACTURING CO. | EASA.IM.R.S.0078 | | FAA STC SR00684SE, DGAC SPAIN | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 BA | Installation of Simplex anti-fire system Model 310-900, 310-1100 and 310-1200 | | | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph #Description#. Compatibility with other aircraft/engine configurations shall be determined by the installer. | 06.10.2004 |
| 10017066 | SIMPLEX MANUFACTURING CO. | EASA.IM.R.S.01502 | | FAA STC SR00889SE | EUROCOPTER FRANCE AS 355 SERIE | Installation of an agricultural spray system. (This STC has been classified as #BASIC# by the FAA in accordance with FAA Order 8110.52 and has been approved using the PTVP Procedures.) | | MDL-210 revision B dated 9 February 2001 or later EASA approved revision # Master Drawing List; 210-102005-001 revision I/R dated 21 December 2000 or later approved EASA revision # Maintenance Manual; 210-102003-002 dated 14 March 2001 or later EASA approved revision # Flight Manual Supplement. | Rotorcraft to be equipped with an EASA approved high skid landing gear; Essential Crew only; Day/Night VFR only; and Airspeed Limitations per Flight Manual Supplement. Prior to installation of this modification the installer must determine t (...) | 26.09.2008 |
| 10043539 | SIMPLEX MANUFACTURING CO. | | REV. 1 | FAA STC SR0227SE | AS 350 B1, AS 350 B2, AS 350 B3, AS 355 E, AS 355 F, AS 355 F1, AS 355 F2, AS 355 N, AS 355 NP | Installation of an airborne high-pressure spray system (Simplex Model 510) capable of cleaning debris from the insulators on electrical power transmission towers. The system consists mainly of an external water tank, a power pack and a turret with spray boom. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | RFMS 510-102003-002, Rev. I/R later revisions of the aboveset documents approved by EASA in accordance with EASA EDD Decision 2004/04/CF (or subsequent revisions of this decision) # and/ or the Technical Implementation Procedures of EU/USA Bilateral (...) | The rotorcraft must have STC EASA.IM.R.S.00622 (Validation of FAA STCSR01289SE) installed to ensure proper operation of the pitot-static instruments. With the Model 510 Spray System installed the rotorcraft is certified for day VFR operations only. With the Model 510 installed no spray or flight operati | 14.02.2013 |
| 10029552 | SINGAPORE TECHNOLOGIES | | | | A320 | Passenger Compartment Reconfiguration | In accordance with the EASA TCDS A.064 plus: The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval-CRI H-01/Special Condition H-01 "Enhanced (...) | A320 LOPA STC Certification Plan, ref. doc. No. 360A023/AED/002, Revision C, dated 31st March 2010 | None | 31.03.2010 |
| 10033242 | SINGAPORE TECHNOLOGIES | | | | A321 | Passenger Compartment Reconfiguration | Certification Basis in accordance with EASA Type Certificate Data Sheets A.064 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. 1. CRI H (...) | - Master Document List, ref. doc. 386/003/AED/002, Rev. A, dated 17th December 2010; LOPA STC Certification Plan, ref. 386/003/AED/001, Rev. B, dated 17th December 2010 or later revisions of the above listed documents approved by EASA | Not applicable | 06.01.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------|----------------|--------------------------|--|--|---|--|---|---------------|
| 10036609 | SINGAPORE TECHNOLOGIES | | | | A320-200 SERIES | Ural Airlines' A320 Cabin Reconfiguration | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:JAR 25 Change 11 and EASA CS 25 Amendments 4 and 10, as per ST Aerospace's Certification (...) | ST Aerospace's Master Document List 399001/AED/002 Revision B, dated 22nd September 2011, or later revisions of the above listed documents approved by EASA | „This design change on Aircraft model Airbus A320-200 is applicable to the following aircraft serial numbers (MSNs): 203; 220; 735; 1152; 1484;1777; 2175; 2167 and 2327. | 22.09.2011 |
| 10045947 | SINGAPORE TECHNOLOGIES | | | | A330-343 | Air Asia X Airbus A330-343 Passenger Compartment Reconfiguration.Certification Plan 395/QAD/001, Air Asia X Airbus A330PassengerCompartment Reconfiguration. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Certification plan: P395/QAD/001, rev J-Master Drawing List: ref 39500000, rev D-Air Asia X A330 Passenger Compartment Reconfiguration Compliance Summary Report: ref P395/QAD/003, rev C-ICA: P395/AED/001, rev - or later revisions of the above listed documents approved by EASA. | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.EE (...) | 31.07.2013 |
| 10035668 | SKY IMD Inc. | | REV. 1 | | SEE APPROVED MODEL LIST | Installation of Sensor Pod and wiring for passenger operated monitoring system | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List (MDL) Document No. PD01 revision H, dated December 13, 2010.Approved Model List (AML) No. SA02248L Revision A, dated February 16, 2011 Instructions for Continued Airworthiness (ICA) Document No. 622 revision E, dated November 15, 20 (...) | This STC is for installation of the sensor pod and wiring from the sensor pod into fuselage. This STC does not include power to the sensor pod or the sensor control unit.Limitations as defined in the applicable Aircraft Flight Manual/Supplements. | 08.07.2011 |
| 10015785 | SKY INTERNATIONAL Inc. | EASA.IM.A.S.01841 | REV 1 | SA10463SC | AVIAT HUSKY A-1B | Installation of a Lycoming model IO-360-A1D6 engine (derated to 196 HP) and MT Propeller model MTV-15-B/205-58 propeller in accordance with Installation Instructions No 1101, Revision C dated 8 May 2006, or later approved revision | | Master Data Drawing List No 1100, Revision C dated 12 May 2006, or later approved revision Installation Instructions No 1101, Revision C dated 8 May 2006, or later approved revision AFMS No 55009-18, Revision IR, dated 8 May 2006, or later approved revision | The engine must be derated to 196 HP (see Installation Instructions for details).The Airplane Flight Manual Supplement No 55009-18, Revision IR or later EASA approved revision, must be used with this modification.The approved Noise Level is 81.6 dB(A). | 23.08.2006 |
| 10035483 | SKYE AVIONICS Ltd. | | | TCCA STC SH11-14 | BELL 407 | Installation of "Check Engine Instruments" Audio Alert System. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | AER-FMS-023 revision 01 dated 29 April 2011 - Flight Manual Supplement or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision)AER-MDL-034 revision NC dat (...) | None | 27.06.2011 |
| 10042298 | SKYTECH, INC. | EASA.IM.A.S.00604 | | FAA STC ST01851NY | PA46-310P | Installation of Piper Mirage PA46-350P Pilot's Heated Windshield into Piper Malibu PA46-310P in accordance with Skytech, Inc. Glass Heated Windshield Installation. HTW-494-782-00, Revision 1 dated December 1, 2003. FAA STC ST01851NY Issued on December 31, 2003. | | - Glass Heated Windshield Installation Manual - Instructions for Continued Airworthiness HTW-494-782-100, dated December 1, 2003. - Airplane Flight Manual Supplement, HTW-494-782-100, Original Issue, dated October 05, 2003. | | 02.03.2005 |
| 10016790 | SKYTRAC SYSTEMS Ltd. | EASA.IM.R.S.01137 | REV. 2 | TCCA STC SH05-42 | BELL 407 EUROCOPTER DEUTSCHLAND BO-105 | SkyTrac ISAT-100 Airborne Position/Data Communicator System Installation.TCCA STC SH05-42 Issue no 1 dated 12 September 2005 | | SkyTrac Systems Inc. Master Document List 250-50097-00 Rev. 01, dated April 13, 2006, or later approved revisions. Supplement Instructions for Continued Airworthiness are provided in SkyTrac Document 270-50097-00 Rev. 00, dated September 1, 2005, or later approved revisions. | 1. Equipment configuration: ISAT-100 (no mod.)/mod A, or later TCCA approved major mod.2. This STC references specific interfacing equipment that is approved via this data.The ISAT-100 equipment is capable of being interfaced to additional aircraft systems for which approval is beyond the | 13.02.2006 |
| 10016798 | SKYTRAC SYSTEMS Ltd. | EASA.IM.R.S.01151 | REV. 3 | TCCA STC SH03-30 | SEE EASA APPROVED MODEL LIST | Installation of SkyTrac ISAT-100/ISAT-200 AIRBORNE POSITION/DATACOMMUNICATOR Revision 1 to add: BK-117 A-1, A-3, B-1, and B-2 models Revision 2 to add: Eurocopter SA 365 C, C1, N, EC 155B Revision 3.1) Adding ISAT-200 Airborne Position/Data Communicato (...) | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition on Lithium Batteries as per EASA CRI F-01 Issue 2. The requirements for environmental protection and the (...) | See EASA Approved Model List. | None | 18.09.2013 |
| 10016927 | SKYTRAC SYSTEMS Ltd. | EASA.IM.R.S.01313 | | TCAA STC SH03-30 ISSUE 5 | EUROCOPTER FRANCE AS 365C EUROCOPTER FRANCE AS 365C1 EUROCOPTER FRANCE AS 365N EUROCOPTER FRANCE EC 155B | Installation of SkyTrac ISAT-100 Airborne Position/ Data Communication System Note: This application is to amend an existing EASA STC (Ref P-EASA.IM.R.S.001151 Rev. 1) to add the SA 365 C, C1, N, EC 155B helicopter models | | - Installation in accordance with Sky Trac System Installation Master Document List 250-50064-00, Rev 07 dated May 16th, 2006, or later approved revisions.-Instructions for Continued Airworthiness provided in SkyTrac Document 270- 50152-00, Rev 00, d (...) | This STC references specific interfacing equipment that is approved via this data. The ISAT 100 equipment is capable of being interfaced to additional aircraft systems for which approval is beyond the scope of this STC. Such interfaces may need fur (...) | 21.03.2007 |
| 10030917 | SKYTRAC SYSTEMS Ltd. | | | TCCA STC SA09-93 | | SkyTrac Flight Following / SATCOM Installation in acc. with TCCA STCSA09-93: ISAT-100 transceiver installed in aft zone of aircraft, Dual element GPS/Iridium antenna installed on upper fuselage, optional DVI-250 and CDP-250 control panels installed at centre radio panel. | The Certification Basis for the original product remains applicable to this certificate/ approval. Additional requirements for the use of Lithium Ion Battery: TCCA Issue Paper IP09074-1, Revision N/C (technically identical with EASA CRI F-56) The requir (...) | Master Drawing List MDL09074, Revision A, dated May 12, 2010 Flight Manual Supplement FMS09074-1, Revision A, TCCA approved May 14, 2010 Instructions for Continued Airworthiness ICA09074-1, Revision A, dated May 11, 2010 or later revisions of the above li (...) | None | 12.07.2010 |
| 10030918 | SKYTRAC SYSTEMS Ltd. | | | TCCA STC SA09-80 | CESSNA 208, 208B | SkyTrac Flight Following / SATCOM Installation in acc. with TCCA STCSA09-80, Issue 1: ISAT-100 transceiver installed in aft zone of aircraft, Dual element GPS/Iridium antenna installed on upper fuselage, optional DVI-250 and CDP-250 control panels installed at centre radio panel. | The Certification Basis for the original product remains applicable to this certificate/ approval. Additional requirements for the use of Lithium Ion Battery: TCCA Issue Paper IP09075-1, Revision N/C (technically identical with EASA CRI F-56) The requir (...) | Master Drawing List MDL09075, Revision A, dated May 11, 2010 Flight Manual Supplement FMS09075-1, Revision A, TCCA approved May 11, 2010 Instructions for Continued Airworthiness ICA09075-1, Revision A, dated April 20, 2010 or later revisions of the above (...) | None | 12.07.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|-------------------|----------------|--------------------------|---|---|---|---|---|---------------|
| 10031247 | SKYTRAC SYSTEMS Ltd. | | REV. 2 | TCCA STC SH10-24 ISSUE 2 | 206A, 206A-1, 206B, 206B-1, 206L, 206L-1, 206L-3, 206L-4, 407 | SkyTrac ISAT Airborne Position/Data Communicator | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | MDL10033 revision N/C dated 10 May 2010 - Master Drawing List: FMS10033-2 revision N/C dated 11 May 2010 - Flight Manual Supplement: CA10033-2, revision N/C dated 11 May 2010 - Instructions for Continued Airworthiness. | STC EASA.IM.R.S.01137 is a prerequisite to this installation. | 22.12.2010 |
| 10031804 | SKYTRAC SYSTEMS Ltd. | | | TCCA STC SA06-85 | 200/200C/200T/200CT/A200/A200C/300/300LW/B300/B300C/1900D/A200CT/B200/B200C/B200T/B200CT | SkyTrac ISAT Position Data Communicator System Installation. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Transport Canada Issue Paper ref IP06103-1 rev NC Special Conditions for Lithium Ion ba (...) | Installation in accordance with Avionics Design Services Ltd Master Drawing List MDL06103 Revision G, dated April 07, 2010. Operation in accordance with SkyTrac Systems Ltd Flight Manual Supplements FMS06103-1 Revision B, FMS06103-2 Revision B and FMS0 (...) | none | 15.09.2010 |
| 10033276 | SKYTRAC SYSTEMS Ltd. | | | TCCA STC SH10-24 ISSUE 2 | EC 120 B | Sky Track ISAT AIRBORNE POSITION / COMMUNICATION UPGRADE Thermofication upgrades an existing ISAT-100 Airborne Position / Data Communicator system installation to an ISAT-200 system (This EASA approval validates the TCCA STC SH10-24, Issue 2) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Special Condition on Lithium Batteries as per EASA CRI F-01 Issue 2. The requirements for (...) | - Avionics Design Services LTD Master Drawing List MDL10033, Revision B, Issue 1, dated December 17, 2010. (relative to the EC 120B) - Avionics Design Services LTD Drawing List DL10033-1, Revision A, Issue 1, dated December 17, 2010. - Flight Manual Sup (...) | None | 11.01.2011 |
| 10033865 | SKYTRAC SYSTEMS Ltd. | | | TCCA STC NO. SH10-24 | EC 135 P1(CDS), EC 135 P1(CPDS), EC 135 P2(CPDS), EC 135 P2+ EC 135 T1(CDS), EC 135 T1(CPDS), EC 135 T2(CPDS), EC 135 T2+ | Sky Track ISAT AIRBORNE POSITION / COMMUNICATION UPGRADE Thermofication upgrades an existing ISAT-100 Airborne Position / Data Communicator system installation to an ISAT-200 system (This EASA approval validates the TCCA STC SH10-24, Issue 2) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Special Condition on Lithium Batteries as per project related EASA CRIF-01 The requirem (...) | - Avionics Design Services LTD Master Drawing List MDL10033, Revision C, Issue 1, dated February 02, 2011 - Avionics Design Services LTD Drawing List DL10033-3, Revision A, Issue 1, dated February 02, 2011 - Flight Manual Supplement FMS10033-3, Revision (...) | none | 15.02.2011 |
| 10036485 | SKYTRAC SYSTEMS Ltd. | | | TCCA STC SH10-24 | AS 355 E, AS 355 F, AS 355 F1, AS 355 F2, AS 355 N | Sky Track ISAT AIRBORNE POSITION / COMMUNICATION UPGRADE Thermofication upgrades an existing ISAT-100 Airborne Position / Data Communicator system installation to an ISAT-200 system (This EASA approval validates the TCCA STC SH10-24, Issue 3 on thetype/models referenced above) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Special Condition on Lithium Batteries as per EASA CRI F-01 Issue 1. The requirements for (...) | - Flight Manual Supplement FMS10033-5, Revision N/C dated March 14, 2011 or later revisions of the above listed documents approved by EASA - Avionics Design Services LTD Master Drawing List MDL10033, Revision D, Issue 1, dated March 1, 2011 (relative t (...) | None | 12.09.2011 |
| 10036492 | SKYTRAC SYSTEMS Ltd. | | | TCCA STC SH10-24 | AS 350 B, AS 350 B1, AS 350 B2 AS 350 B3, AS 350 BA, AS 350 D | Sky Track ISAT AIRBORNE POSITION / COMMUNICATION UPGRADE Thermofication upgrades an existing ISAT-100 Airborne Position / Data Communicator system installation to an ISAT-200 system (This EASA approval validates the TCCA STC SH10-24, Issue 3 on thetype/models referenced above) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Special Condition on Lithium Batteries as per EASA CRI F-01 Issue 1. The requirements for (...) | - Flight Manual Supplement FMS10033-4, Revision N/C dated March 14, 2011 or later revisions of the above listed documents approved by EASA - Avionics Design Services LTD Master Drawing List MDL10033, Revision D, Issue 1, dated March 1, 2011 (relative to (...) | None | 13.09.2011 |
| 10041706 | SKYTRAC SYSTEMS Ltd. | | | TCCA STC SH10-51 | AW139 | Installation of a SkyTrac ISAT-100/ISAT-200 Position Data Communicator (Canadian STC SH10-51 issue 2) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List, document HP 139-23-00-104 Revision A, dated May 11, 2012 and Transport Canada approved May 28, 2012 Rotorcraft Flight Manual Supplement, document FMS HP 139-23-301 Revision A, dated October 7, 2011 and Transport Canada approved May (...) | None | 09.10.2012 |
| 10014444 | SLINGSBY ADVANCED COMPOSITES LTD. | EASA.A.S.02495 | REV. 1 | | BEECH 76 | Introduction of Bendix/King KCS 55A Compass, KX165 NAV/COM, KR87 ADF, King KN62A DME & KI229 Systems | | Singsby Aviation Ltd Modification SAL/MOD/BE76/01 Singby Aviation Ltd AFM Supplement Singby Beech 76-01 Issue 1 or later approved revision, Singby Aviation Ltd AFM Supplement Singby Beech 76-02 Issue 1 or later approved revision. | | 20.06.2006 |
| 10013975 | SMA - SOCIETE DE MOTORISATIONS | EASA.A.S.00774 | REV. 7 | | 1. 182M, 182N, 182P, 182Q, 182R 2. F182-P, F182-Q | -DET C-09-503 Air Supply circuit modification DET C-08-508 Introduction of new powerplant limitations-DET C-07-521 Kelly Aerospace alternator ES7024-16 Installation- DET C-08-510 Modification of the intercooler inlet and outlet nozzle geometry -DJC-S (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TR-C182-AIM-01-C08508TR-C182-AIPCS-01-C08508TR-C182-ASMS-01-C08508TR-C182M-AFMS-01-E-C08508TR-C182M-AFMS-01-F-C08508TR-C182N-AFMS-01-E-C08508TR-C182N-AFMS-01-F-C08508TR-C182PQR-AFMS-01-E-C08508TR-C182PQR-AFMS-01-F-C08508 or later revisions of the abo (...) | Applicable to Cessna 182R, Q, P, N, M and Reims Aviation F182Q and P modified in accordance with STC SMA.EASA.A.S.00774 and later approved revisions. | 04.05.2011 |
| 10016495 | SMA ENGINES INC. | EASA.IM.A.S.02857 | | FAA STC SA03302AT | CESSNA 182Q CESSNA 182R | Installation of Societe de Motorisations Aeronautiques (SMA) Aircraft Diesel Engine (ADE) model SR305-230-1 (or SR305-230 converted to model SR305-230-1 by incorporation of SMA Service Bulletin SB-01-76-002), MT propeller model MTV-9-BS/198-58B propeller and other minor modifications. | | Installation in accordance with SMA Engines Inc Installation Instructions "SMA Engines Inc Airplane installation Manual for Cessna 182 fitted with an SMA SR305-230-1 Engine" Rev Basic Issue dated June 30, 2006 and Master Drawing List SP01170314-0 Rev (...) | None. | 06.11.2008 |
| 10015728 | SNACKY CREEK ENTERPRISES, LLC | EASA.IM.A.S.01745 | REV 1 | FAA STC SA02177CH | BAE JETSTREAM 3201 SERIES | Removal of passenger seats and installation of Spectrum Aeromed medical systems. Note: This revision reflects the transfer of this STC from Spectrum Aeromed, Inc to Snacky Creek Enterprises, LLC. | | Master Document List 6034-1A, Revision 1, dated July 6, 2005, Spectrum Aeromed Inc AFM supplement 6034-1FM Spectrum Aeromed Instructions for Continued Airworthiness Manual 6043-1M or later revision of the above listed documents approved by EASA in ac (...) | | 07.01.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|-------------------|----------------|---------------------------------|--|--|---|--|--|---------------|
| 10016347 | SNAKY CREEK ENTERPRISES, LLC | EASA.IM.A.S.02664 | | FAA STC ST01248CH; LBA STC TA0 | BAE 125 SERIES 1000A BAE 125 SERIES 1000B HAWKER 1000 | Installation of an air ambulance conversionNote:1. This STC reflects the transfer of LBA STC TA0328 from Spectrum Aeromed, Inc to Snaky Creek Enterprises, LLC.2. This STC supersedes LBA STC TA0328. | | Spectrum Aeromed Inc. Drawing 6036-DMS, No Revision, dated January 26, 2000 or later revision of the above listed document approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision). | AFM Supplement dated August 4, 2007 or later approved revision is required | 07.01.2008 |
| 10016348 | SNAKY CREEK ENTERPRISES, LLC | EASA.IM.A.S.02666 | | FAA STC ST01607CH; LBA STC TA0 | LEARJET 31 LEARJET 31A LEARJET 35 LEARJET 35A | Installation of an air ambulance conversionNote:1. This STC reflects the transfer of LBA STC TA0565 from Spectrum Aeromed, Inc to Snaky Creek Enterprises, LLC.2. This STC supersedes LBA STC TA0565. | | Spectrum Aeromed Inc. Drawing 6017-DMS-Z1, Revision 2, dated December 2, 2001 or later revision of the above listed documents approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision). | 1. AFM Supplement dated December 7, 2001 or later approved revision is required.2. Installation requires previous or concurrent installation per LBA STC TA0512 (STC ST01026V1). | 07.01.2008 |
| 10016349 | SNAKY CREEK ENTERPRISES, LLC | EASA.IM.A.S.02668 | | FAA STC SA00882CH; LBA STC SA | BEECH 200 BEECH 200C BEECH B200 BEECH B300 BEECH B300C | Installation of Spectrum Aeromed medical systems. | | Spectrum Aeromed Inc drawing 6014-DMS, Revision -, dated 02/04/1998FMS6014, dated 05/08/1998Spectrum Aeromed Inc drawing 60108-DMS-1, Revision IR, dated 04/05/2002FMS6008-DMS-1, dated 09/11/2002or later revision of the above listed document approved (...) | | 07.01.2008 |
| 10016350 | SNAKY CREEK ENTERPRISES, LLC | EASA.IM.A.S.02669 | | FAA STC SA00972CH; LBA STC SA | PIPER PA-42 PIPER PA-42-1000 PIPER PA-42-720 | Installation of an air ambulance conversion | | Spectrum Aeromed Inc. Drawing 6004, Revision 1, dated March 9, 1998, or later revision of the above listed document approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision). | | 07.01.2008 |
| 10016988 | SNAKY CREEK ENTERPRISES, LLC | EASA.IM.R.S.01394 | | FAA STC SR176CH; DGAC FRANCE S | BELL 212 BELL 412 | Installation of an air ambulance unitNote:1. This STC reflects the transfer of DGAC France STC SR176CH from Spectrum Aeromed, Inc to Snaky Creek Enterprises, LLC.2. This STC supersedes DGAC France STC SR176CH. | | Spectrum Aeromed Report No. SI2055A, dated May 20, 1993 or later approved revision. | | 07.01.2008 |
| 10035259 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STC SA02935CH | CESSNA 441 | Air Ambulance Conversion in accordance with Spectrum Aeromed DrawingList | The Certification Basis of the original product remains applicable to this change. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA approved Master Drawing List, Doc. No. 6078DL, Rev. -, dated 16-Feb-2010 (>-<-) FAA Approved Aircraft Flight Manual Supplement, Doc. No. 6078FMS, Rev. -, dated 12-Jul-2010 or later revisions of the above listed documents approved by EASA in accordance with (...) | None | 09.06.2011 |
| 10035308 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STC SA02897CH | 510 | Air Ambulance Conversion according FAA STC SA02897CH. | The Certification Basis for the original product remains applicable to this certificate/ approval. CS 23 as defined in CRI A-01 and indicated in TCDS EASA.IM.A.502. The requirements for environmental protection and the associated certificated noise and/ (...) | Installed law. Spectrum Aeromed Master Drawing List 6053DL, dated 21 Apr 2010. Operated law. FAA approved AFM-S 6053FMS, dated 26 Oct 2010. Maintained law. FAA accepted Inst. for ContAW, Manual 6053MM, dated 21 Apr 2010 or later revisions of the above listed (...) | According limitation section of AFM-S. | 14.06.2011 |
| 10035962 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STC SA02930CH | CESSNA 208, 208B | Air Ambulance Conversion in accordance with Spectrum Aeromed DrawingList | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA approved Master Drawing List, Doc. No. 6006DL, Rev. 3, dated 24-Jan-2011 (>-<-) FAA Approved Aircraft Flight Manual Supplement, Doc. No. 6006FMS, Rev. -, dated 30-Nov-2010 or later revisions of the above listed documents approved by EASA in accordance with (...) | None | 01.08.2011 |
| 10036411 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STCSA01213CH; CAA ES S374-I | SEE EASA APPROVED MODEL LIST | Install air ambulanceNote:1. This STC reflects the transfer of STC CAA ES S374-I from Spectrum Aeromed, Inc to Snaky Creek Enterprises, LLC.2. This STC supersedes STC CAA ES S374-I, EASA STC EASA.IM.A.S.C.010163. This STC, together with EASA STCs 10036412 and 10036413 supersedes EASA STC EASA.IM.A.S.024 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Attached MDL or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 26.07.2012 |
| 10036412 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STCSA01213CH; CAA ES S374-I | SEE EASA APPROVED MODEL LIST | Install air ambulanceNote:1. This STC reflects the transfer of STC CAA ES S374-I from Spectrum Aeromed, Inc to Snaky Creek Enterprises, LLC.2. This STC supersedes STC CAA ES S374-I, EASA STC EASA.IM.A.S.C.010163. This STC, together with EASA STCs 10036411 and 10036413 supersedes EASA STC EASA.IM.A.S.024 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Attached MDL or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 26.07.2012 |
| 10036413 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STCSA01213CH; CAA ES S374-I | SEE EASA APPROVED MODEL LIST | Install air ambulanceNote:1. This STC reflects the transfer of STC CAA ES S374-I from Spectrum Aeromed, Inc to Snaky Creek Enterprises, LLC.2. This STC supersedes STC CAA ES S374-I, EASA STC EASA.IM.A.S.C.010163. This STC, together with EASA STCs 10036411 and 10036412 supersedes EASA STC EASA.IM.A.S.024 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Attached MDL or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 26.07.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|--------------|----------------|----------------------|---|---|--|--|------------|---------------|
| 10039407 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STC ST02068CH | BAE.125 SERIES 1000A BAE.125 SERIES 1000B BAE.125 SERIES 800A BAE.125 SERIES 800B BH.125 SERIES 400A BH.125 SERIES 600A DH.125 SERIES 400A HAWKER 1000 HAWKER 800 HAWKER 800XP HAWKER 850XP HAWKER 900XP HS.125 SERIES 400B HS.125 SERIES 400B/1 HS.125 SERIES 401B HS.125 SERIES 403A(C) HS.125 SERIES 403B HS.125 SERIES 600A HS.125 SERIES 600B HS.125 SERIES 600B/1 HS.125 SERIES 600B/2 HS.125 SERIES 600B/3 HS.125 SERIES 700A HS.125 SERIES 700B HS.125 SERIES F400B HS.125 SERIES F403B HS.125 SERIES F600B | To Install an Air Ambulance Conversion.This Change instructs the interior and electrical changes to theircraft. Stretcher and medical equipment installation will be certified by the appropriate vendor. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Installed in accordance with Spectrum Aeromed Inc Master Document List6033-2DL, Revision 2, dated August 26, 2010;Operation in accordance with Spectrum Aeromed Inc Approved FlightManualSupplement 6033-2FMS, Revision 2, dated September 15, 2011;Mainta (...) | None. | 02.05.2012 |
| 10039716 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STC SR02982CH | AW109SP | EASA Validation of FAA STC SR02982CH "Installation of equipment railsand mounts in accordance with RSG AeroDesign Master Document List00A0216002, No Revision, dated February 23, 2011, or later FAA approvedrevisions" | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | RSG AeroDesign Master Document List 00A02160002, No Revision, datedFebruary 23, 2011<->->FAA approved RFM Supplement 19A0210001, dated June 15, 2011<->->FAA accepted Instructions for Continued Airworthiness, Document No.16A02160002, NoRevision, (...) | None | 18.05.2012 |
| 10039717 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STC SR02981CH | AW109SP | EASA Validation of FAA STC SR02981CH "Installation of a medical seat,fire extinguisher, utility track and cylinder mount in accordance withRSG AeroDesign Master Document List 00A02390001, No Revision, datedFebruary 23, 2011, or later FAA approved revisions" | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | RSG AeroDesign Master Document List 00A02390001, No Revision, datedFebruary 23, 2011<->->FAA approved RFM Supplement 19A02390001, dated June 15, 2011<->->FAA accepted Instructions for Continued Airworthiness, Document No.16A02390001, No Revision (...) | None | 18.05.2012 |
| 10039718 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STC SR02974CH | AW109SP | EASA Validation of FAA STC SR02974CH "Air Ambulance Conversion inaccordance with Spectrum Aeromed Master Document List 7011DL, Revision1, dated May 18, 2011, or later FAA approvedrevisions" | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Spectrum Aeromed Master Document List 7011DL, Revision 1, dated May 18, 2011<->->FAA approved Flight Manual Supplement7011 FMS<->-> Revision 1 datedApril 17, 2012<->->FAA accepted Instructions for Continued Airworthiness, Manual 7011MM, NoRevis (...) | None | 18.05.2012 |
| 10039719 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STC SR02978CH | AW109SP | EASA Validation of FAA STC SR02978CH "Installation of shorepowermechanical provisions in accordance with RSG AeroDesign Master DocumentList 00A02240001, No Revision, dated February 23, 2011, or later FAAApproved revisions" | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | RSG AeroDesign Master Document List 00A02240001, No Revision, datedFebruary 23, 2011<->->FAA accepted Instructions for Continued Airworthiness, Document No.16A02240001, No Revision, dated April 19, 2011for later revisions of the above listed docume (...) | None | 18.05.2012 |
| 10039720 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STC SR02977CH | AW109SP | EASA Validation of FAA STC SR02977CH "Installation of a medical floorliner in accordance with RSG AeroDesign Master Document List00A02160003, No Revision, dated February 23, 2011, or later FAA approvedrevisions" | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | RSG AeroDesign Master Document List 00A02160003, No Revision, datedFebruary 23, 2011<->->FAA accepted Instructions for Continued Airworthiness, Document No.16A02160003, No Revision, dated April 19, 2011for later revisions of the above listed docume (...) | None | 04.06.2012 |
| 10039721 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STC SR02980CH | AW109SP | EASA Validation of FAA STC SR02980CH "Installation of an interior panelsystem in accordance with RSG AeroDesign Master Document List00A02160001, No Revision, dated February 23, 2011, or later FAA approvedrevisions" | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | RSG AeroDesign Master Document List 00A02160001, No Revision, datedFebruary 23, 2011<->->FAA accepted Instructions for Continued Airworthiness, Document No.16A02160001, No Revision, dated April 19, 2011for later revisions of the above listed docume (...) | None | 18.05.2012 |
| 10041098 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STC SA02488CH | PC-12 PC-12/45 PC-12/47 | Air Ambulance Conversioninstall air ambulance conversion inaccordance with Spectrum AeromedInc. Master Drawing List 6045DL, dated July 23, 2007 or later FAAapproved revisions. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | FAA approved Aircraft Flight Manual Supplement, 6045FMS, dated September21, 2007;FAA accepted Instructions for Continued Airworthiness, Manual 6045MM,dated May 30, 2007;or later revisions of the above listed documents approved by EASA inaccordance w (...) | None. | 21.08.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|-------------------|----------------|----------------------|---|---|--|---|--|---------------|
| 10042970 | SNAKY CREEK ENTERPRISES, LLC | LBA STC TA0637 | REV. 1 | FAA STC ST01612CH | CL-600-1A11 (CL-600) CL-600-2A12 (CL-601) CL-600-2B16 (CL-601-3A) CL-601-3R, CL-604 CL-600-2B19 | Air Ambulance Conversion | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC No. ST01612CH dated 20 March 2013. Spectrum Aeromed Inc. Master Drawing List 6041DL, Revision 7, dated 20 August 2012. Spectrum Aeromed Inc. Flight Manual Supplement 6041 FMS, Revision 7 dated 16 October 2012. Spectrum Aeromed Inc. Instructions f (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 23.04.2013 |
| 10043060 | SNAKY CREEK ENTERPRISES, LLC | EASA.IM.A.S.03094 | | FAA STC SA00560CH | PA-31P, PA-31T, PA-31T1, PA-31T2, PA-31T3, PA-31P-350 | Air Ambulance Conversion. | | - Installation must be in accordance with Master Drawing List 2003DL Rev.1, dated 2 April 2008 or later approved revisions. - Operation must be in accordance with Aircraft Flight Manual Supplement 2003FMS Rev.1 (->-) or later approved revision. - Maint (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The ins (...) | 18.06.2009 |
| 10043384 | SNAKY CREEK ENTERPRISES, LLC | | | FAA STC SA02488CH | PC-12, PC-12/45, PC-12/47 PC-12/47E | Air ambulance conversion in accordance FAA STC SA02488CH. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Aircraft Flight Manual Supplement, Spectrum Aeromed Manual 6045FMS, Initial Release, dated September 21, 2007. Instructions for Continued Airworthiness, Spectrum Aeromed Manual 6045MM, Initial Release, dated May 30, 2007; or later revisions of the above (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 25.01.2013 |
| 10014506 | SOCATA | EASA.A.S.02576 | | | TBM 700 | STC 34.001.06/Garmin GNS530 TAWS. | | Garmin Document 190-00181-00 Rev. E Flight Test Report TBM-00-REV-188/11 December 2007 Supplement to FM ref. STC 34-001-06 Certification Report. | TBM 700 equipped Garmin GNS 530 only s/n to 433 excepted 269. | 30.01.2008 |
| 10014524 | SOCATA | EASA.A.S.02607 | | | TB 200 | Validation of definition specificities of TB200 S/N 1633 aircraft, compared to TB200 GT certified aircraft definition | | TB200 S/N 1633 specific supplement to basic TB200 Pilot Operating Handbook | STC 53.001.06 applicable to TB200 s/n 1633 only | 20.07.2006 |
| 10015065 | SOCATA | EASA.A.S.03391 | | | SOCATA TBM 700 | Installation of a multi-functions display Garmin GMX 200 connected to already installed equipments (weather radar, COM/NAV/GPS, Traffic Advisory System). | | Definition and Installation: EADS SOCATA Modification Data Sheet (MDS) n° STC.34.001.07, edition 1 revision 0, dated 15/01/2008 or later EASA approved revisions. Operations: EADS SOCATA TBM 700 Pilot Operation Handbook (POH) Supplement n° 46A/Garmin GM (...) | This revision can be installed on TBM 700 s/n to 433 when equipped with EADS SOCATA optional Modifications n° OPT 70-23-019, OPT 70-23-018, OPT 70-34-047 and OPT 70-34-040. The installation can be performed only by EADS SOCATA technical staff according to the MDS here-above referred. | 06.02.2008 |
| 10029243 | SOCATA | P-EASA.A.S.03962 | REV. 4 | | TB 20 TB 21 | Installation of Garmin G500 EFIS Interfaced with a dual Garmin GNS430W/G530W communication and navigation system, for IFR B-RNAV/P-RNAV operations with coupling to autopilot. Revision 3 of the STC including the extent to RNP LNAV and LPV approach operations (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 23 The Certified noise and/ or emissions levels of the original product are unchanged (...) | Installation Instructions n° STC 34.001.09.010, Edition 1 Revision 0 dated 05/10/2012. Instructions for continued airworthiness n° STC 34.001.09.006 Edition 1 Revision 0, dated 05/01/2010. LRU software versions listing n° STC 34.001.09.009 Edition 3 Revis (...) | The modified airplane must be equipped with an altitude indicator, an airspeed indicator, an altitude indicator, amagnetic heading indicator and a radionavigation indicator independent from the G500 EFIS (->-) a third radionavigation or radio commu (...) | 23.10.2012 |
| 10040655 | SOCATA | | | | DHC-6 SERIES 300 | Avionics upgrade based on Garmin G600 EFIS and GNS 430 W system, according to SOCATA modification file n° STC.00.001.11 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 23 Standard CRIs B52, F51, F52, F54, F58, F84 The certified noise and/ or emission (...) | Socata Flight Manual Supplement n° 38, Issue 1 dated 15 March 2012. Socata Maintenance Manual Supplement revision 0 dated 15 February 2012. Socata Modification Sheet n° STC.001001.11, Edition 1 dated 4 July 2012 or later revisions of the above listed documents approved by EASA | This design change is limited to airplanes s/n 292, 298, 300, 730 and 745. Initial installation must be performed by authorized Socata technical personnel. This avionics installation is approved for IFR operations in BRNAV, PRNAV, RNP approach LNAV and LP (...) | 17.07.2012 |
| 10042727 | SOCATA | | | | TB20, TB21 | Installation of TAS Garmin GT5800 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Garmin Instructions for Continued Airworthiness for GTS 8XX Traffic System, revision 2 dated 29 October 2010. Socata Flight Manual Supplement 32, Edition 1 Revision 2 dated 15 November 2012, English version p/n Z00.182S23T0E1R2 Socata Flight Manual Supple (...) | This STC is not applicable to airplanes equipped with a 14 V electrical system. Initial installation of this design change must be performed by authorized SOCATA technical personnel. Modified airplanes must be equipped with Socata modification No.34-001-009 approv (...) | 14.12.2012 |
| 10028245 | SOCATA SAS | P-EASA.A.S.03961 | REV. 2 | | TBM 700 N | Installation of a Honeywell HF KHF 1050 communication system on aircraft TBM 700 N equipped with Garmin 1000 Glass Cockpit System - Update for CA inclusion | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Daher-Socata A/S Instructions for Continued Airworthiness ref.Z00.DMNSTC01PEE0R0, original issue dated July 2010. Daher-Socata A/S Modification File No STC-23.001.09, edition 1, dated 09/12/2009 or later revisions of the above listed documents approved by EASA | None | 26.08.2010 |
| 10036696 | SOCATA SAS | | | | AS 355 F2 | Installation of Provisions for WESCAM MX-10 Camera installation | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Design Change Description for Mission System - DAHER SOCATA No STC.25.001.11 Edition 1 revision 1 dated 28 May 2011 | Serial Numbers 5026, 5299, 5298, 5156 only | 30.09.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|----------------|----------------|----------------------|---|--|---------------------|---|--|---------------|
| 10013926 | SOCIETE AIR FRANCE S.A. | EASA.A.S.00198 | REV. 1 | | BOEING 747-400 | Air France B747-400 "ULYSSE" cabin reconfiguration. This is a revision to STC Ref EASA.A.S.00198, to extend the applicability to the aircraft B747-400 F-GITH MSN 32868, F-GITI MSN32869 and F-GEXA MSN 24154. The general description is included in the Air France Letter BL.NI 10 002/EB and in the AEO | | | 1.The approval of this modification is limited to the Boeing B747-400 MAS 24969, 24990, 25344, 25600, 25601, 25602, 32868, 32869, 32871, 25238, 25302, 25559, 25628, 25630, 24154, 24155. The limitations and conditions defined in the EASA Datasheet ar (...) | 19.10.2005 |
| 10013927 | SOCIETE AIR FRANCE S.A. | EASA.A.S.00225 | REV. 2 | | BOEING 747-400 | ISPPSS Installation on B747-400 in accordance with MA.PM 263/03 rev 5 dated 22/09/2005 | | AMM Supplement MAMP 482/03PD rev 01 dated 28/01/05AFM Supplement to DGU10002-428 dated 01/09/04 and AFM Supplement DGU10002-4B3 dated 01/09/04 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph #Description#. Compatibility with other aircraft/engineconfigurations shall be determined by installer. | 20.02.2006 |
| 10013939 | SOCIETE AIR FRANCE S.A. | EASA.A.S.00375 | REV. 1 | | AIRBUS A319-115 | A319 CJ CASTA II Cabin Reconfiguration | | Change Approval Sheet CAS-A320-04-002 rev 01 dated February1, 2005Air France Industries Airplane Flight Manual Supplement (AFM-S) ref.: "Government of France, F-RBFA / F-RBFB", Revision 02, dated February 1, 2005, or later EASA-approved revision | 1. This STC is limited to the A319-115 CJ MSN's 1485 and15562. This STC is approved only for the product configuration as defined in the approved design data referred to inthe paragraph "Description". Compatibility with other aircraft/engine c (...) | 18.10.2005 |
| 10013947 | SOCIETE AIR FRANCE S.A. | EASA.A.S.00405 | REV 1 | | BOEING 777-200-228-300 | RAAS (Runway Awareness and Advisory System) activation. AirFrance AEO 77-34-0051 rev. 2 dated November 24, 2004CAS-B777-05-19 rev 0 dated 29 June 2006 (or later approved issue)RAAS Activation on B777-200/-300 Air France aircraftNote: Thismodification (...) | | DCS1W001 Rev 01 for B777-200D631W002 Rev 01 for B777-300 | 1. See AFM limitations for options to be used.2. Applicable to a/c serial numbers listed in Air France CAS-B777-05-0192. This STC is approved only for the product configuration as defined in the approved design data referred to inthe paragr (...) | 13.07.2006 |
| 10013996 | SOCIETE AIR FRANCE S.A. | EASA.A.S.01030 | REV 1 | | AIRBUS A321-211 | Cabin reconfigurationChange Approval Sheet ref. CAS-A320-04-015 revision 2, dated October 21, 2005 | | Change Approval Sheet ref. CAS-A320-04-015 revision 0, dated October 21, 2005 | This STC is limited to the A321 MSN 1017 | 08.09.2006 |
| 10014054 | SOCIETE AIR FRANCE S.A. | EASA.A.S.01137 | | | BOEING 767-200-300 | Installation of Elementary Surveillance Dual ATC Mode S on B767-300The ATC Mode S computers TPR 720 are replaced by a TPR 901 which is compliant with ATC Mode S Elementary surveillance requirements.ATC Mode S Elementary surveillance implementation i (...) | | Air France Change Approval Sheet No. CAS-B767-04-001 dated December 1, 2004.Air France Service Bulletin No. AF SB B767-04-001-34-001 #EATC Installation on B767-300# | | 22.08.2005 |
| 10014055 | SOCIETE AIR FRANCE S.A. | EASA.A.S.01138 | | | BOEING 767-200-300 | Installation of Elementary Surveillance Dual ATC Mode S on B767-300The ATC Mode S computers TPR 720 are replaced by a TPR 901 which is compliant with ATC Mode S Elementary surveillance requirements.ATC Mode S Elementary surveillance implementation i (...) | | Air France Change Approval Sheet No. CAS-B767-04-001 dated December 1, 2004.Air France Service Bulletin No. AF SB B767-04-001-34-001 #EATC Installation on B767-300# | | 07.04.2005 |
| 10014056 | SOCIETE AIR FRANCE S.A. | EASA.A.S.01139 | | | BOEING 747-200-300 | Installation of EGPWS. This STC supersedes and cancels DGAC STC IM29-SF-0282. | | Group 1 (Air France Fleet): AEO 47-34-0715.Group 2 (Customer Fleet): CAS B747-04-010.AFM Supplements D6-33747-228, D6-33747-283 and D6-33747-383 are applicable to Air France B747-200/-300 aircraft.AFM Supplement No. MA.PM0006/05 is applicable to Pulmantur Air aircraft. | | 18.05.2005 |
| 10014057 | SOCIETE AIR FRANCE S.A. | EASA.A.S.01140 | | | BOEING 747-300 | Elementary surveillance dual ATC Mode S installation. | | Change Approval Sheet ref. CAS B747-04-009 Revision 0, dated February 10, 2005 or later approved revision. | This STC is limited to the B747 MSN 24106. | 15.03.2007 |
| 10014065 | SOCIETE AIR FRANCE S.A. | EASA.A.S.01151 | | | AIRBUS A320 | Cabin passenger seats relocation and floor proximity emergency escape path marking change | | Change approval sheet No CAS-A320-05-001 revision 1, dated March 22, 2005 | | 07.04.2005 |
| 10014337 | SOCIETE AIR FRANCE S.A. | EASA.A.S.01594 | | | AIRBUS A330-203 MSN 422, 443, 448, 458, 465, MSN 498, 500, 502, 503, 516, | Air France A330 "Ulysse" cabin reconfiguration in accordance with Change Approval Sheet No. CAS-A330-04-001 dated 16/09/2005 (or later EASA approved revision) | | | The approval of this STC is limited to Airbus A330-203 Serial Numbers 422, 443, 448, 458, 465, 481, 498, 500, 502, 503,516, 519. The Limitations and conditions defined in the EASA Datasheet are applicable.A copy of this STC and the CAS must be enclosed in the files of each modified aircraft. | 12.10.2005 |
| 10014338 | SOCIETE AIR FRANCE S.A. | EASA.A.S.01596 | | | BOEING 747-400 | Air France B747-400 Big Bins modification. This is a revision to the EASA STC Ref 2004-4528, to extend the applicability to Boeing747-400, F-GEXA, MSN 24154.The general description is included in the Air France Letter BL.NI 10 003/EB and in the AEO (...) | | | 1. The approval of this modification is limited to the Boeing B747-400 MSN 24969, 24990, 25344, 25600, 25601, 25602, 32868, 32869, 32871, 25238, 25302, 25559, 25628, 25630, 24154,24155.The limitations and conditions defined in the EASA Datasheet are (...) | 19.10.2005 |
| 10014375 | SOCIETE AIR FRANCE S.A. | EASA.A.S.01650 | | | BOEING 747-200 BOEING 747-300 | Suppression of climb inhibition above flight level 300 on TCAS computer in accordance with AEO 47-34-0714 Revision 1 dated 17/10/2005 | | AEO 47-34-0714 Revision 1 dated 17/10/2005 | | 23.11.2005 |
| 10014399 | SOCIETE AIR FRANCE S.A. | EASA.A.S.01693 | REV. 1 | | BOEING 767 | Cabin Reconfiguration from Eva Air to Aeroflot Standard | | Air France Modification CAS-B767-05-007 | FAA STC SA5950NM (Door 2 Aft Galley Installation) is not included in this approval. | 17.07.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|----------------|----------------|----------------------|----------------------------------|---|---------------------|---|--|---------------|
| 10014451 | SOCIETE AIR FRANCE S.A. | EASA.A.S.02504 | | | BOEING 737-800 BOEING 737-900 | KLM 737-800, -900 Recaro Seats Installation | | - Certification Programme CP-B737-06-001 Rev2, March 28th 2006- AFBS B737-06-001-25-001 Rev 00, March 24th 2006-04-06 | - Compatibility of this design change with other previously approved modifications must be determined by the installer. - The installation is limited to the B 737-800, -900 listed in the CP-B737-06-001 Rev2, March 28th 2006. - As indicated in the doc. (...) | 11.05.2006 |
| 10014454 | SOCIETE AIR FRANCE S.A. | EASA.A.S.02506 | | | BOEING 767-300 | Cabin Reconfiguration | | Air France Change Approval Sheet No EA33-B767-06-001 rev 0 | -- | 26.06.2006 |
| 10014456 | SOCIETE AIR FRANCE S.A. | EASA.A.S.02510 | | | AIRBUS A330-343 | Cabin Reconfiguration with the Installation of new B/C Seating Configuration as per Air France Change Approval Sheet n°CAS A330-05-012 | | - Type Design Document Ref. MF-A330-05-012- Supplements to instructions for Continued Airworthiness and Airplane Flight Manual according to Modification Airworthiness Section of Doc. Ref. MF-A330-05-012- Air France SB A330-05-012-25-001- Air France SB A330-05-012-25-002 | | 30.04.2007 |
| 10014463 | SOCIETE AIR FRANCE S.A. | EASA.A.S.02524 | | | BOEING 777-200 | Air France B777-200 "Ulysse" cabin modification with Contour (CAS B777-04-003) or BEA business class seats (CAS B777-05-015). | | Air France Modification B777-04-003Air France service bulletinsB777-04-003-25-001 rev 0 dated February 14 2006B777-04-003-25-002 rev 0 dated February 14 2006B777-04-003-25-003 rev0 dated February 14 2006or later approved revisionsThe Air France Modi (...) | Limited to Aircraft serial numbers as defined in Air France document CAS B777-04-003 and CAS B777-05-015 | 03.11.2006 |
| 10014465 | SOCIETE AIR FRANCE S.A. | EASA.A.S.02526 | | | AIRBUS A320-211 | Photo-luminescent EEPM system installation | | Change Approval Sheet (CAS), ref. A320-06-004 revision , dated May 24, 2006 or later EASA approved revisionAirplane Flight Manual Supplement (AFM-S), ref. MA.PM 017/06 CE revision0, dated February 3, 2006 or later EASA approved revision | This STC is limited to the A320 MSN's 211, 215 and 270 | 06.06.2006 |
| 10014501 | SOCIETE AIR FRANCE S.A. | EASA.A.S.02568 | | | A320 | Cabin Reconfiguration in bi-classes on LAN A320 | | AFSB A320-06-006-25-001 Rev.00MF A320-06-006 Rev.00 or later approved revision | 1. Limited to installation on MSN 1304, 1332, 1351 and 1355.2. Compatibility of this design change with other previously approved modifications must be determined by the installer. | 30.10.2006 |
| 10014531 | SOCIETE AIR FRANCE S.A. | EASA.A.S.02617 | | | BOEING 737-300 BOEING 737-500 | ATC Mode S-Enhanced Surveillance-Provision and Activation on Boeing 737-300/-50 in accordance with the Modification File MF-B737-06-004, Rev 00, dated 11 October 2006. | | - Modification File MF-B737-06-004, Rev 00, dated 11 October 2006. - AFMS-B737-06-004-001 First Issue dated 3 May 2006, or later approved revision. - AFMS-B737-06-004-002 First Issuedated 9 May 2006, or later approved revision. - ICA-B737-06-004 Rev 01, or later approved revision. | 1. Limited to installation on Boeing 737-300/-500.2. Compatibility of this design change with other previously approved modifications must be determined by the installer. | 23.10.2006 |
| 10014570 | SOCIETE AIR FRANCE S.A. | EASA.A.S.02674 | | | BOEING 767 | B767 Cabin Reconfiguration from KLM to Aeroflot. | | CAS-B767-06-008 | | 20.07.2007 |
| 10014611 | SOCIETE AIR FRANCE S.A. | EASA.A.S.02738 | | | AIRBUS A340-300 | THY A340 Cabin Retrofit Modification | | Air France Change Approval Sheet n° CAS A340-06-007 Rev. 0 | | 16.05.2007 |
| 10014663 | SOCIETE AIR FRANCE S.A. | EASA.A.S.02820 | REV 1 | | BOEING 737-300 BOEING 737-500 | Installation of elementary surveillance Dual ATC Mode S on B737-300/-500 in accordance with the Modification File MDF-B737-04-004, Rev 01, dated 11 October 2006. | | - Modification Design File MDF-B737-04-004, Rev 01, dated 11th October 2006. - Modification Airworthiness File MAF-B737-04-004 Rev 00, or later approved revisions. - AFBS B737-04-004-34-002 , first Issue- ICA-B737-04-004 Rev 00, or later approved revision. | 1. Limited to installation on Boeing 737-300/-500.2. Compatibility of this design change with other previously approved modifications must be determined by the installer. | 23.10.2006 |
| 10014719 | SOCIETE AIR FRANCE S.A. | EASA.A.S.02896 | | | BOEING 757-200 | Installation of ATC Mode S Elementary and Enhanced Surveillance | | Modification Definition File, MF-B757-06-001, Rev 00, Date23/03/2007AFM Supplement, AFMS-B757-06-001, Rev 00, 15/01/2007Instructions for Continued Airworthiness, ICA-B757-06-001, Rev 00, 03/01/2007 | Certification basis as FAA TCDS A2NMincluding:TGL 13 Rev 1,01/06/2003EASA AMC-20.13, 02/12/2005JAA TGM 21/07, 01/10/2003. | 28.03.2007 |
| 10014822 | SOCIETE AIR FRANCE S.A. | EASA.A.S.03034 | | | BOEING 767-300 | Installation of Elementary Surveillance Dual ATS Mode S. | | Air France Change Reference CAS-B767-04-001 Rev 1. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 26.02.2007 |
| 10014865 | SOCIETE AIR FRANCE S.A. | EASA.A.S.03089 | | | AIRBUS A340-300 | A340 FINNAIR Cabin and IFE Upgrade | | 2. Certification Program ref. doc. CP-A340-07-005, Rev. 3, dated 15th July 2008.2. Modification File, ref. doc. MF-A340-07-005, Rev. 0 dated 17th September 2008.2. Change Approval Sheet, ref. doc. CAS-A340-07-005, dated 17th September 2008.2. 1 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 18.09.2008 |
| 10015195 | SOCIETE AIR FRANCE S.A. | EASA.A.S.03581 | REV 1 | | BOEING 777-200 BOEING 777-300 | ADS-B Out Activation. | | AFMS-B777-07-023CAS B777 07 023 rev0 11Jul08 | None. | 05.09.2008 |
| 10015245 | SOCIETE AIR FRANCE S.A. | EASA.A.S.03681 | | | BOEING 747-400 | ADS-B Out Installation and Activation as per Air France Service Bulletin B747-08-004-34-001 rev. 0 or later approved revisions. | | AFMS B747-08-004 Airplane Flight Manual Supplement for B747-400 ADS-B Out Installation. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 10.12.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|------------------|----------------|----------------------|---|--|---|---|--|---------------|
| 10027295 | SOCIETE AIR FRANCE S.A. | P-EASA.A.S.03975 | REV. 2 | | A320 | Cabin Reconfiguration from Dragon Aviation Leasing to Juneyao Standard (DKH) Revision of the EASA STC 10027295 to remove EWIS limitation | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- CRI H-01/Special Condition H-01 "Enhanced Airworthiness (...) | - Certification Program, ref. doc. No. CP-A320-09-009, Revision 02, dated 31st August 2009- Statement of Compliance, ref. doc. No. SOCP-A320-09-009-005, dated 31st August 2009- Instruction for Continued Airworthiness, ref. doc. No. ICA-A320-09-009 Rev (...) | None | 23.11.2010 |
| 10032363 | SOCIETE AIR FRANCE S.A. | | | | A340-211, A340-212 | A340 Royal Jordanian - New IFE and Cabin reconfiguration This STC consists in a cabin reconfiguration by installing new economic class seat and a new In-Flight Entertainment (IFE) system (Sigma Aero IFE SIT). | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- CS 25 Amendment 8- Special Condition H-01 : Enhanced Airworthiness Program for Aeropla (...) | Change Approval Sheet, Ref : CAS-A340-09-009 Rev 00, dated 29 October 2010 Air France Service Bulletin, Ref : AFSB-A340-09-009-25-001 or later revisions of the above listed documents approved by EASA | This STC is applicable to the following aircraft MSNs :-, MSN 14,-, MSN 22,-, MSN 38,-, MSN 43 | 29.10.2010 |
| 10038157 | SOCIETE AIR FRANCE S.A. | | | | 777-200 | Alitalia Cabin Reconfiguration | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- CRI H-1 ICA on EWIS, CRI E-10 Inflatable Seat Belts | Change approval sheet CAS-B777-11-003 Rev. 0 dtd. 19.Jan 12 or later revisions of the above listed documents approved by EASA | 777-200 MSN as defined in Change approval sheet CAS-B777-11-003 | 24.01.2012 |
| 10042281 | SOCIETE AIR FRANCE S.A. | | | | 747-400 | CorsairFly cabin reconfiguration and SIT IFE installation, Cabin refurbishment and new IFE system installation. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval- CRI D-GEN02 PTC (HR/SD seat materials) CRI E-22 (Cabin Attendant Direct View) The requir (...) | CAS-B747-11-012 revision 0, dated 20th November 2012 - Change Approval Sheet SOCP-B747-11-012-037 rev 1, dated 15th November 2012 - Instructions for Continued Airworthiness or later revisions of the above listed documents approved by EASA | none | 21.11.2012 |
| 10042639 | SOCIETE AIR FRANCE S.A. | | | | A319-111, A319-113, A320-211, A320-212, A321-111, A321-212. | Installation of the ACARS (Aircraft Communication Addressing and Reporting System) CMU-900 to replace the Teletype MU45B | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - Certification Program CP-A320-12-003 Rev. 2, dated 03/10/2012 - Change Approval Sheet CAS-A320-12-003 Rev. 0, dated 19/10/2012- Airplane Flight Manual Supplement AFMS-A320-12-003-001 Rev.0, dated 23/07/2012- Compliance Check List CCL-A320-12-003 Rev (...) | This approval is applicable to the following MSN: 61 ; 63 ; 102 ; 115 ; 144 ; 145 ; 155 ; 196 ; 187 ; 214 ; 227 ; 236 ; 237 ; 239 ; 270 ; 278 ; 285 ; 286 ; 287 ; 337 ; 352 ; 377 ; 491 ; 498 ; 509 ; 521 ; 529 ; 544 ; 598 ; 600 ; 608 ; 618 ; 625 ; 637 ; (...) | 12.12.2012 |
| 10014926 | SOCIETE D'ETUDE ET DE GESTION | EASA.A.S.03176 | | | BEECH 200 | Installation of a support base for a stretcher. | | Installation in accordance with SEGA Modification ref A-25-02-07-TwinJet, latest approved revision. Operation in accordance with SEGA Supplement Manuel de Vol Beech 200 ref FAM No A-25-02-07-TwinJet, dated May 2007, or later approved revision. | | 08.10.2007 |
| 10014927 | SOCIETE D'ETUDE ET DE GESTION | EASA.A.S.03177 | | | BEECH 90 | Installation of a support base for a stretcher. | | Installation in accordance with SEGA Modification ref A-25-02-07-TwinJet, latest approved revision. Operation in accordance with SEGA Supplement Manuel de Vol Beech 90 ref. FAM No A-02-02-07-TwinJet dated May 2007, or later approved revision. | | 08.10.2007 |
| 10015255 | SOCIETE D'ETUDE ET DE GESTION | EASA.A.S.03707 | | | BEECH 1900 | Installation of a galley. | | Description: #Installation sur avion Beech 1900 d'une Armoire Self-Service# Ref.: Modification A-25-01-08-TwinJet P-EASA.A.S.03707 Edition 0 dated 13 January 2009 Master Document List P-EASA.A.S.03707 Ref.: Modification A-25-01-08-TwinJet MDol Editio (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 16.01.2009 |
| 10015294 | SOCIETE D'ETUDE ET DE GESTION | EASA.A.S.03838 | | | PIPER PA-34 | Pneumatic deicer removal. | | SEGA application document no. A-03-09-Deice DA edition 20 dated 20 March 2009 SEGA Flight Manual Supplement no. A-03-09-Deice edition 0 dated 20 March 2009 SEGA modification file no. A-03-09-Deice edition 0 dated 20 March 2009 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 21.04.2009 |
| 10017182 | SOCIETE D'ETUDE ET DE GESTION | EASA.R.S.01315 | | | EUROCOPTER FRANCE AS 355F1 | Installation of rack and sending/reception antennae. | | H-25-02-06-LIVE TOOLS Annex 2 dated May 2008 or later approved revision # Dossier Vol de Controle H-25-02-06-LIVE TOOLS Annex 3 Edition 0 dated 21 May 2008 or later approved revision # Supplement Manuel de Vol H-25-02-06-LIVE TOOLS Annex 4 date May 20 (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr (...) | 12.06.2008 |
| 10017237 | SOCIETE D'ETUDE ET DE GESTION | EASA.R.S.01423 | | | EUROCOPTER FRANCE AS 350 | Modification H-25-04-07-SFP - installation of deployable antenna mounted on landing gear. | | Definition, installation, and inspection & maintenance: INSTALLATION SUR HELICOPTERE ECUREUIL AS350D#UNE ANTENNE DE RETRANSMISSION D#IMAGES MODIFICATION H-25-04-07-SFP; Dossier de Modification - Janvier 2008; Annexe 1 - Dossier Justificatif - J (...) | This STC provides approval for the installation and deployment of the antenna. A separate approval is required for the installation and operation of any equipment transmitting through the antenna. The installation of this modification by third persons (...) | 25.09.2008 |
| 10017238 | SOCIETE D'ETUDE ET DE GESTION | EASA.R.S.01424 | | | EUROCOPTER FRANCE AS 355 SERIE | Modification H-25-04-07-SFP - installation of deployable antenna mounted on landing gear. | | Definition, installation, and inspection & maintenance: INSTALLATION SUR HELICOPTERE ECUREUIL AS355D#UNE ANTENNE DE RETRANSMISSION D#IMAGES MODIFICATION H-25-04-07-SFP; Dossier de Modification - Janvier 2008; Annexe 1 - Dossier Justificatif - J (...) | This STC provides approval for the installation and deployment of the antenna. A separate approval is required for the installation and operation of any equipment transmitting through the antenna. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 25.09.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------|------------------|----------------|----------------------|---|---|--|---|---|---------------|
| 10027064 | SOCIETE D'ETUDE ET DE GESTION | P-EASA.A.S.03925 | REV. 1 | | MS893 A, B, E, E-D MS894 A, C, E RALLYE 235 A, C, E, E-D, F | Reduction of maximum take off and landing weights to 1000 kg. | FAR 23. | SEGA Flight Manual Supplement A-05-09-1000kg, ed. 0, dtd. 20-07-2009, ENand FR version. SEGA Installations Instructions A-05-09-1000kg, ed. 0, dtd. 20-07-2009. SEGA Modification File A-05-09-1000kg, ed. 0, dtd. 20-07-2009. | MS894 models: seating capacity limited to 3 persons, pilot included. Rallye 235 models: seating capacity limited to 3 persons, pilot included. Fuel tanks capacity limited to 240 l (2*120). The airplane configurations listed in attachment 1 are exclude (...) | 08.01.2010 |
| 10027989 | SOCIETE D'ETUDE ET DE GESTION | P-EASA.A.S.03973 | | | DHC-8-100, -200, -300 | Installation of ELT KANNAD 406 AF-COMPACT or ARTEX ME406 | CS 25 Amendment 6 as defined in 21A.101(a) has been used instead of TCDSEASA.I.M.A.191The Certification Basis for the original product remains applicable tothis certificate/ approvalThe certificated noise and/ or emissions levels of the original prod (...) | A-BALISE406-DHC8; KANNAD DOC08038C; ARTEX 570-1600 Rev. Gor later revisions of the above listed documents approved by EASA | None | 24.11.2009 |
| 10028678 | SOCIETE D'ETUDE ET DE GESTION | EASA.R.S.01561 | | | EUROCOPTER FRANCE AS 350 SERIE | | | | | 31.07.2009 |
| 10028679 | SOCIETE D'ETUDE ET DE GESTION | EASA.R.S.01560 | | | EUROCOPTER FRANCE AS 355 SERIE | | | | | 31.07.2009 |
| 10028814 | SOCIETE D'ETUDE ET DE GESTION | P-EASA.R.S.01492 | REV. 3 | | 1. AS350 B, AS350 B1, AS350 B2 1. AS350 B3, AS350 BA 1. AS350 D, AS350 BB 2. AS355 E, AS355 F, AS355 F1 2. AS355 F2, AS355 N, AS355 NP | Modifications H-25-05-06-SFP and H-25-05-06-SFPant - Installation of arack and transmitting/receiving antennaeNotes:Initial issue of this STC covered installation of a rack - modificationH-25-05-06-SFPRevision 1 to this STC added installation of tra (...) | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | H-25-05-06-SFP Flight Manual Supplement - RFMS for Installation of anElectronic Rack for Reception/Sending onEurocopter AS350/AS355 (Ed. dtdtd 10 Dec 2009 - document in English)H-25-05-06-SFPant Flight Manual Supplement - RFMS for Installation of Re (...) | This STC is not applicable to AS350B3 model helicopters withmodification OP-4305 (Arriel 2D engine installation) incorporated. | 26.07.2012 |
| 10029480 | SOCIETE D'ETUDE ET DE GESTION | | | | AS350 B2 | Installation system for "EC120 type" seats on Eurocopter AS350B2 -modification H-08-09-Siège | As per EASA TCDS.R.008 Issue 03, 12 October 2007.The Certification Basis for the original product remains applicable tothis certificate/ approval.The certificated noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | Installation system for EC120 seats on Eurocopter AS350(Modification H-08-09-Siège - Ed. 0 dated February 25th 2010)Application document H-08-09-Siège_DA - Ed.0 - February 25th 2010 | None. | 26.03.2010 |
| 10032781 | SOCIETE D'ETUDE ET DE GESTION | | | | DR300/108, DR300/120, DR300/125, DR300/140, DR300/180 R, DR315, DR340, DR360, DR380, DR400/100, DR400/120 D, DR400/125, DR400/120, DR400/120 A, DR400/140, DR400/140 B, DR400/160, DR400/160 D, DR400/180 S, DR400/200 R, DR400/180, DR400/180 R, DR400/2+2, DR400/500, DR400/NGL | A-07-09-Filtre Ed. 0 - Installation of remote of filter kit | The Certification Basis for the original product remains applicable tothis certificate/ approval.The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | Document d'Application A-07-09-Filtre_DA du 15 Novembre 2010or later revisions of the above listed documents approved by EASA | None | 01.12.2010 |
| 10036788 | SOCIETE D'ETUDE ET DE GESTION | | REV. 1 | | 1900 SERIES | Installation of a partition in the rear cabin area. | The Certification Basis for the original product remains applicable tothis certificate/ approval. The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Installation in accordance with SEGA Modification File A-25-Cloison_DA,dated September 2011;Instructions for continued airworthiness as defined in SEGA documentA-25-Cloison<->->-dated September 2011; or later revisions of the above listeddocuments approved by EASA. | None. | 03.01.2012 |
| 10038783 | SOCIETE D'ETUDE ET DE GESTION | | | | 1. AS350B/BB/BA/DB1/B2 2. AS350B3 (SEE CONDITIONS) 3. AS355E/F/F1/F2/N/NP | Front Lower Cowling Modification (Reference H-01-11-Capot)Separation of cowling in two halves reassembled by fasteners.Note: This STC is also applicable to front lower cowlings modified asper STC EASA.R.S.01359). | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Dossier de Modification - Front Lower Cowling ModificationonEurocopter Ecureuil Ref. Modification H-01-11-Capot, Ed. 1 - dated 08March 2012"Document d'Application" Ref. A-01-11-Capot_DA, Ed. 1 - dated 08 March 2012 | None | 14.03.2012 |
| 10039483 | SOCIETE D'ETUDE ET DE GESTION | EASA.R.S.01391 | REV. 1 | | 1- AS350B, AS350BA 1- AS350B3, AS350D 1- AS350BB, AS350B1, AS350B2 2- AS355E, AS355F, AS355F1 2- AS355F2, AS355N, AS355NP | Installation of an Outside Aerial Picture System.Revision 1adds installation of ALIMOUNT III camera system. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | SUPPLEMENT MANUEL DE VOL - INSTALLATION SUPPORTS SYSTEME DECAMERA -AS350B, BA, BB, B1, B2, B3, D, AS355E, F, F1, F2, N, NP (Modification:H-25-03-07-Aile Image), Ed.0 R1 / Janvier2012; or later revisions of the above listed documents approved by EASA. (...) | This STC does not approve Installation of an Outside AerialPictureSystem on AS350B3 helicopters incorporating modification OP-4305(Arriel2D engine installation). Compliance against the requirements of applicable certification basis for AS350B3 helicopters incorporatingOP4305 hasn't been verified. | 07.05.2012 |
| 10039562 | SOCIETE D'ETUDE ET DE GESTION | | REV. 1 | | 1900 | Installation of a galley in accordance with SOCIETE D'ETUDEET DEGESTION AERONAUTIQUE (SEGA) Master Document ListA-11-03-TwinJet-GalleyII_MDoL Edition 1 April 2012 | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Installation Procedure: Installation sur avion Hawker Beechcraft 1900d'une armoire self-service, ref. SEGA A-11-03-TwinJet-GalleyII_DA, Edition 1, April 2012<->->-Aircraft Flight Manual Supplement: Supplement manuel de vol, ref. SEGA A-11-03-TwinJet (...) | None | 05.06.2012 |
| 10038292 | SOFINARE | | | | R182 TR182 | Installation of integrated Flight System Garmin G-500 and DoubleNav/Com/GPS systems GNS-430W. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval:CS-23, incl. Amdt. 2 (Corrigendum) | AFMS (GNS-430W); SOFINARE 21J.081.0821.016a, Initial IssueAFMS (G500); SOFINARE 21J.081.0821.016b, Initial Issuor later revisions of the above listed documents approved by EASA. Continued Airworthiness Instructions Sofinare 21J.081.0821.010.00, EASA (...) | None. | 03.02.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------|----------------|----------------|----------------------|-------------------------|--|--|---|---|---------------|
| 1003948 | SOFINARE | | | | 1900D | INSTALLATION OF TERRAIN AWARENESS AND WARNING SYSTEM (TAWS). | The Certification Basis for the original product remains applicable to this certificate/ approval. | Technical Documents to be Approved by EASA: -, SOFINARE 21J.081.0869.003 Definition Document -, SOFINARE 21J.081.0869.016AFM Supplementor later revisions of the above listed documents approved by EASA -, SOFINARE 21J.081.0869.010 Continuous Airworthine (...) | None (Applicable to All Models Beechcraft 1900D Airplanes) | 04.06.2012 |
| 10013935 | SOGERMA SERVICES | EASA.A.S.0034 | REV. 1 | | AIRBUS A310-304 | Aircraft modification for 8.33 kHz spacing on VHF system in accordance with Modification Approval Sheet SOG 23-878 Issue 1 dated 01 July 2004 or later EASA-approved revision. Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services. | | | | 07.01.2008 |
| 10014149 | SOGERMA SERVICES | EASA.A.S.0127 | REV 1 | | AIRBUS A320-211 | Cabin layout change from 144 pax to 155 pax configuration in accordance with the Master Drawing List ref. MDL-25-861, issue B, dated August 6, 2004 or later EASA approved revision. Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services. | | | | 07.01.2008 |
| 10014307 | SOGERMA SERVICES | EASA.A.S.01551 | REV 1 | | ATR 42-320 | Additional probes in RAL configuration in accordance with MAS SOG-01-459 issue 1 dated 30/11/05. Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services. | | AFM supplement Ref 240/04/SOG/DEV issue 4 | 1. MSN 078 Associated STC No. EASA.A.S.01219 Revision 1 | 04.01.2008 |
| 10014348 | SOGERMA SERVICES | EASA.A.S.01612 | REV 1 | | AIRBUS A340-642 | This STC deals with the installation of 4 new double seats in Y class (Kyoto manufacturer). Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services. | | The associated Modification Approval Sheet is referenced SOG-25-1233 issue 1 dated 30/11/05. All documents related to this STC are listed in the Master Data List Drawing n° MDL-25-1233, Issue No B, (or later EASA approved revision) TECHNICAL REPERCUSSION SHEET reference 21-1233 issue B. | 1. This STC is approved only to allow airline customer to install on a trial basis a few seats on one aircraft (MSN 706). 2. The limitations and conditions defined in the EASA Datasheet are applicable. A copy of this STC and of the MAS/MDL must (...) | 04.01.2008 |
| 10014365 | SOGERMA SERVICES | EASA.A.S.01635 | REV 1 | | A340-642 | This STC deals with the installation of a 2 new triple seats in W class (Contour design seats). Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services. | | The associated Modification Approval Sheet is referenced SOG-25-1217 issue 1 dated 30/11/2005. All documents related to this STC are listed in the Master Data List Drawing n° MDL-25-1217, Issue No B, (or later EASA approved revision) TECHNICAL REPERCUSSION SHEET reference 25-1217 issue B. | 1. This STC is approved only to allow airline customer to install on a trial basis a few seats on one aircraft (MSN 706). 2. The limitations and conditions defined in the EASA Datasheet are applicable. A copy of this STC and of the MAS/MDL must (...) | 04.01.2008 |
| 10014416 | SOGERMA SERVICES | EASA.A.S.0173 | REFV 1 | | AIRBUS A340-642 | Installation in upper class side facing Britax Malcom seats equipped with the inflatable restraints, monuments, a bar and a chandelier according to MAS SOG-25-597 issue 4 (dated 02/07/04) and reference documents: Airplane Flight Manual A340-600 1.0 (...) | | | | 07.01.2008 |
| 10014428 | SOGERMA SERVICES | EASA.A.S.02474 | REV 1 | | DC-9-21 | Modification Janus Phase I: Cancellation of the restriction of flights in icing conditions. Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services. | | Modification Approval Sheet (MAS), ref. SOG-00-092, issue 3 and dated April 12, 2005 Airplane Flight Manual Supplement, ref. BD/1251/01, issue 17, dated April 11, 2005 | 1. This STC is limited to the MSN 47308. | 04.01.2008 |
| 10014476 | SOGERMA SERVICES | EASA.A.S.02538 | REV 1 | | AIRBUS A320-211 | Cabin modification from 155 to 167 Y/C. Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services. | | Modification Approval Sheet (MAS) ref. SOG-25-1250 issue 1, dated April 14, 2006 or later EASA approved revision | 1. This STC is limited to the A320 MSN 615. | 04.01.2008 |
| 10014862 | SOGERMA SERVICES | EASA.A.S.03086 | | | AIRBUS A300 MODEL B2-1C | Replacement of the obsolete RCA weather radar by a Honeywell RDR-4B colour display radar system with predictive windshear function. | | # Modification Approval Sheet, ref: SOG-34-2000 Issue 1, dated 10 May 2007# Aircraft Flight Manual Supplement, ref: AFM-SOG-A0-34-2000 Issue B | 1. This STC is limited to the MSN 0032. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other (...) | 29.05.2007 |
| 10014899 | SOGERMA SERVICES | EASA.A.S.03129 | | | A310-325 | Installation of ATC mode S Enhanced Surveillance (EHS) This Supplement Type Certificate consists in the installation of new ATC EHS in order to be compliant with the new ATC mode S Enhanced Surveillance regulation. | | Modification Approval Sheet, ref: SOG-34-2005 issue 1, dated 13 July 2007 AFM Supplement, ref: AFM-SOG-A1-34-2005 issue B, dated 02 May 2007 Instruction for operation: Master Minimum Equipment List Supplement | 1. This STC is limited to A310-325 aircraft model: MSN 636- MSN 642. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification (...) | 19.07.2007 |
| 10015013 | SOGERMA SERVICES | EASA.A.S.03315 | | | AIRBUS A340-313 | Virgin Atlantic A340 Cabin Modification. | | Modification Approval Sheet, ref. S-00-2020, issue 1, dated 10.06.2008 Flight Manual Supplement, ref. FMS-F4027-00-01, dated 23.05.2008 Instruction for Continued Airworthiness, ref. ICA-F4027-00-01, issue D, dated 02.06.2008 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 12.06.2008 |
| 10015078 | SOGERMA SERVICES | EASA.A.S.03412 | | | AIRBUS A320 SERIES | Installation of passenger entertainment cabin audio system on seats. Note: This STC supersedes DGAC France STC 180-SF-0175. | | Service Bulletin SOG-DO-23-33-01 | | 04.01.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------|---------------------|----------------|--------------------------------|--|--|--|---|--|---------------|
| 10015079 | SOGERMA SERVICES | EASA.A.S.03413 | | DGAC FRANCE STC 180-SF-0174 | AIRBUS A320 SERIES | Sony Video System InstallationNote: This STC supersedes DGACFrance STC 180-SF-0174. | | Service Bulletin SOG-DO-23-36-01 | | 04.01.2008 |
| 10015080 | SOGERMA SERVICES | EASA.A.S.03414 | | DGAC FRANCE STC 072-MJ-0857 | AIRBUS A300 MODEL B2-1A | Installation of a GPS/HIS adapterNote: This STC supersedes DGAC France STC 072-MJ-0857. | | SOG-34-500 issue C | | 04.01.2008 |
| 10015081 | SOGERMA SERVICES | EASA.A.S.03415 | | DGAC STC F-C180-SF-0242; EASA | AIRBUS A320 SERIES | Cabin conversion modification *20 B/C - 148 Y/C*Note: This STC supersedes DGAC France STC F-C180-SF-0242 and EASA 2004-9168. | | Modification Approval Sheet (MAS) referenced SOG-25-882 issue 2, dated June 25, 2004 | | 04.01.2008 |
| 10015082 | SOGERMA SERVICES | EASA.A.S.03416 | | DGAC FRANCE STC F-C072MJ0864; | AIRBUS A300-600 | Installation of an ELT 96 emergency locator transmitterNote: This STC supersedes DGAC France STC F-C072MJ0864 and EASA 2004-4595. | | Modification Approval Sheet SOG-25-070 issue D, dated 31 March 2004 | | 04.01.2008 |
| 10015083 | SOGERMA SERVICES | EASA.A.S.03417 | | DGAC FRANCE STC 180-SF-0229; E | AIRBUS A320 SERIES | Cabin conversion from 180 Y/C to 16 B/C + 169 Y/C*Note: This STC supersedes DGAC France 180-SF-0229 and EASA 2004-3515. | | Modification Approval Sheet (MAS) SOG-25-796 issue 2, dated 12 March 2004 | | 04.01.2008 |
| 10015084 | SOGERMA SERVICES | EASA.A.S.03418 | | DGAC FRANCE STC 180-SF-0221; E | AIRBUS A320 SERIES | Installation of BENNET seat containerNote: This STC supersedes DGAC France 180-SF-0221 and EASA 2004-2036. | | Modification Approval Sheet (MAS) SOG-25-755 issue 2, dated 10 February 2004AFM Supplement ref. AFM-MD050-25-755-01 | | 04.01.2008 |
| 10015085 | SOGERMA SERVICES | EASA.A.S.03419 | | DGAC FRANCE STC 072-MJ-0862; E | AIRBUS A300 MODEL B2 | SPC antenna modificationNote: This STC supersedes DGAC France STC 072-MJ-0862 and EASA 2003-304. | | SOG-05-340 issue E | 1. Valid only for Airbus B2 MSN 003 | 04.01.2008 |
| 10015086 | SOGERMA SERVICES | EASA.A.S.03420 | | DGAC FRANCE STC 072-MJ-0861; E | AIRBUS A300 MODEL B2 | Cargo door modification for use of SPC antennaNote: This STC supersedes DGAC France STC 072-MJ-0861 and EASA 2003-303. | | Modification Approval Sheet (MAS) SOG-52-340 issue E | 1. Valid only for MSN 003. | 04.01.2008 |
| 10015087 | SOGERMA SERVICES | EASA.A.S.03421 | | EASA 2004-2363 | AIRBUS A300 MODEL B2 | Installation of a ground proximity warning systemNote: This STC supersedes EASA STC 2004-2363. | | MAS N SOG-34-576 Issue 3AFMS DIT/M/G/2881/03 | 1. Only valid for MSN 003. | 04.01.2008 |
| 10015208 | SOGERMA SERVICES | EASA.A.S.03623 | | | DHC-6-300 | S-53-2039 Structural Provision for Instrument Panel. | | MAS S-53-2039 Issue 3 dated 26 August 2008. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 27.08.2008 |
| 10043103 | SOGERMA SERVICES | EASA.A.S.0168 | REV. 2 | | A310-304, MSN 537 AND 586 | Traffic Alert and Collision Avoidance System, TCAS change 7 software and two ATC Mode S installations in accordance with Modification Approval Sheet SOG 34-879 Issue 2 dated 08 July 2004 | | | 1 This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description" Compatibility with other aircraft/engine configuration shall be determined by the installer this certificate shall (...) | 08.09.2004 |
| 10016846 | SOLOY, LLC | EASA.IM.R.S.01222 | | FAA STC SR01647SE | EUROCOPTER FRANCE AS 350 B2 | Installation of Honeywell LTS101-700D-2 engine. | | See attached Annex. | See attached Annex. | 20.03.2008 |
| 10027209 | SOLOY, LLC | P-EASA.IM.A.S.03096 | | | CESSNA 206H, CESSNA T206H | Installation of Rolls-Royce 250-B17F/2 series engine in a Cessna 206H/T206H. | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the certificated noise and/or emissions levels. | Installation of Rolls-Royce 250-B17F/2 series engine with a Hartzell Model HC-B3TF-7A/T10173FN(K)-11R Propeller in accordance with Soloy Master Drawing List 766-100, Rev. 0, dated October 2008. Operation must be in accordance with Aircraft Flight Manual (...) | None | 14.09.2009 |
| 10028355 | SOLOY, LLC | | | FAA STC SA01985SE | 206H, T206H | Installation of New Rear Observation Seat in Cessna 206 | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation of new rear observation seat in accordance with Soloy, LLC Master Drawing List 910-100, Rev E, dated March 26, 2009 and Service Instruction No. 910-01, dated March 19, 2009 or later revisions of the above listed documents approved by EASA (...) | Weight and Balance: Equipment Removal Item, Weight (lb), Station (in) Seat, 2nd Row (2 Seats), 57, 76.0 Seat, 3rd Row (Bench), 36, 105.2 Equipment Installed Item, Weight (lb), Station (in) Seat Assembly with Armrests, 34.0, 81.7 Base Plate Assembly, 23.2, 81 (...) | 21.12.2009 |
| 10028448 | SOLOY, LLC | P-EASA.IM.A.S.03237 | | FAA STC SA01688SE | 206H, T206H | Installation of a revised shoulder harness restraint system and new rear cabin headliner in a Cessna 206H/T206H | FAR 23 as defined in FAA TCDS A4CEThe Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation of a revised shoulder harness restraint system and new rear cabin headliner in accordance with Soloy, LLC Master Drawing List 912-100, dated May 31, 2008 (-/-) Maintenance in accordance with Soloy, LLC "Instructions for Continuous Airworthiness (...) | N/A | 12.01.2010 |
| 10028449 | SOLOY, LLC | P-EASA.IM.A.S.03236 | | FAA STC SA00858SE | 206, U206B, U206C, U206D TU206A, U206E, U206F TU206D, TU206E, U206, TU206F U206A, TU206G, 206H, T206H U206F, TU206C, U206G | Installation of Observation Window in Cessna 206 | FAR 23 as defined in FAA TCDS A4CEThe Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation of observation window in accordance with Soloy, LLC Master Drawing List, 901-100, Rev. B, dated November 15, 2000, and manufacturing order for Part 901-1011, dated December 6, 2000 or later revisions of the above listed documents approved by (...) | As per FAA STC SA00858SE, "This modification has been determined to be compatible with Soloy STC SA2353NM and airplanes in landplane, floatplane, and skiplane configuration when modified in accordance with FAA approved data." Installer must check airtail (...) | 12.01.2010 |
| 10028758 | SOLOY, LLC | P-EASA.IM.R.S.01457 | | FAA STC SR00805SE | AS350BA | Installation of Honeywell LTS101-600A-3A engine on Eurocopter AS350BA helicopter. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Modifications completed prior to 31 January 2007: EMDA-199-B revision 1 dated 16 February 2005 - Installation of Honeywell engine model LTS 101-600A-3A; EDHA-197-B revision 2 dated 18 February 2005 - Maintenance manual Supplement. Modifications complete (...) | See attached Data Sheet | 14.06.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------|----------------|----------------------|--|---|---|--|--|---------------|
| 10046687 | SOLOY, LLC | | | FAA STC SA02271SE | A) TU206G, U206G B) 206H, T206H | Fabrication and installation of Soly wing mounted LED landing and taxilight kits 909-1000 in accordance with FAA STC SA02271SE, reissued April 9, 2013. | FAR-23, IR., 23.303, 23.307, 23.333, 23.335, 23.337, 23.341, 23.609, 23.615, 23.625, 23.627, 23.1351(a)(b)(c)(d), 23.1367(a)(b)(c)(d), 23.1363, 23.1581, 23.1583(h) FAR-23, Amdt. 5., 23.1585(a) FAR 23, Amdt. 7., 23.611 FAR-23, Amdt. 14., 23.1365 FAR-23, Am (...) | Pilot's Operating Handbook and Airplane Flight Manual Supplement (AFMS), Doc. No. S2090A, Revision 0, FAA approved March 26, 2013, Master Drawing List (MDL), Doc. No. 909-100, Revision C, dated May 15, 2013, Instructions for Continued Airworthiness (ICA) (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 11.10.2013 |
| 10015900 | SOUTHEAST JET GROUP, INC. | EASA.IM.A.S.02021 | | ST00589W1 | CESSNA 501, CITIATION CESSNA 550 CITIATION II CESSNA 551, CITIATION III CESSNA S550, (CITIATION S/II) | Installation of Honeywell EDZ-605 Two-Tube Electronic Flight Instrumentation System (EFIS). | | Columbia Avionics, Inc. Master Document List No. 313330014, Revision 2, dated November 27, 1996 or later approved revisions. Columbia Avionics, Inc. Document No. 305430005, FAA approved Airplane Flight Manual Supplement, dated October 24, 1997 of (...) | | 24.08.2006 |
| 10016786 | SPACECAM SYSTEMS INC. | EASA.IM.R.S.01128 | REV. 1 | TCCA STC SH02-6 | AS 350 B AS 350 B1 AS 350 B2 AS 350 B3 AS 350 BA AS 350 D AS 355 E AS 355 F AS 355 F1 AS 355 F2 AS 355 N | Spacecam RAMS Installation | | Definition and Installation: DECA Aviation Modification Summary MS01071, revision 3, June 6th 2005 or later Transport Canada (TCCA) approved revisions. DECA Aviation Installation instructions II 01071, revision 2, June 6 2005 or later Transport Canada (...) | Eurocopter (Aerospatiale) Service Bulletin N° 53-05 and 53-06 (for Tail Boom Balast) are required as part of this installation. This STC is the EASA Validation of Transport Canada STC n° SH02-6 Issue 2, dated June 28, 2005 | 25.08.2009 |
| 10014403 | SR TECHNICS AIRCRAFT DESIGN SERVICE | EASA.A.S.01698 | | | BOEING 737-33V | Reconfiguration of six Boeing 737-33V from 149TC to 16BC/100TC including installation of G2, G7 and LAV E | | SR Technics (Ir) documents C737-25-1024-CCS, Rev 3 C737-25-1024-MDL-01, Rev 3 | B737-33V Serial Numbers 29337 to 29342 inclusive | 16.03.2006 |
| 10014694 | SR TECHNICS AIRCRAFT DESIGN SERVICE | EASA.A.S.02860 | | | BOEING 747-400 SERIES | Reconfiguration of five Boeing B747-400 aircraft to 54J/62W/228Y. | | C747-25-1002-CCS, Revision 0, or later approved revision C747-25-1002-MDL, Revision 0, or later approved revision | 1) Limited to the following registration Mark/Serial Number: aircraft: G-VFAB/24958, G-VHOT/26326, G-VBIG/26255, G-VVOW/32745, G-VROC/32746. 2) The following EASA approved STC's must be installed concurrently with this design change: EASA.A.S.03165E, EASA.IM.A.S.02572 | 08.10.2007 |
| 10014007 | SR TECHNICS SWITZERLAND Ltd. | EASA.A.S.01047 | REV. 1 | | B747-400 | Reconfiguration of Virgin Atlantic Airways B747-400 from LHR to LGW configuration in accordance with SR Technics Master Data List C747-25-1001-MDL-01, Rev 3 or later approved revisions | The Certification Basis for the original product remains applicable to this certificate/ approval. | C747-25-1001-CCS, Rev 3, C747-25-1001-MDL-01, Rev 3 or later revisions of the above listed documents approved by EASA | none | 24.02.2011 |
| 10014568 | SR TECHNICS SWITZERLAND Ltd. | EASA.A.S.02672 | REV. 1 | | A319-112 | In-Flight telecommunication equipment - antenna installation (partial removal of GTE Airfone System installed by FAA STC ST00129NY-Dand SA1693GL) The Revision 1 was issued to reflect the transfer from SR Technics Ireland to SR Technics Switzerland LTD | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | GTE Airfone STC Documentation Checklist: AFD70-0016-00, Rev. 3 SR Technics Doc.: C320-23-1005-CCS-01 Rev. 0 or later revisions of the above listed documents approved by EASA | MSN 1077, 1084, 1089, 1136, 1147, 1155, 1170, 1209, 1277 | 10.02.2011 |
| 10029069 | SR TECHNICS SWITZERLAND Ltd. | | REV. 1 | | B747-400 | Reconfiguration of Virgin Atlantic Airlines B747-400 Aft Upper Deck in accordance with SR Technics Master Data List C747-25-1005-MDL Rev. 0, or later approved revisions | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The applicant elected to comply to FAR Part 25 at amendment 128 exceptor the followin (...) | C747-25-1005-ICA-01 Instructions for Continued Airworthiness Rev. 0 C747-25-1005-WBR-01 Weight and Balance report Rev. 0 or later revisions of the above listed documents approved by EASA | This STC is limited to the following A/C: 1., B747-4Q8, G-VFAB, Tab RT945, MSN 24958. 2., B747-4Q8, G-VHOT, Tab RT946, MSN 26326. 3., B747-4Q8, G-VBIG, Tab RT947, MSN 26255. 4., B747-41R, G-VVOW, Tab RM052, MSN 32745. 5., B747-41R, G-VROC, Tab (...) | 24.02.2011 |
| 10031834 | SR TECHNICS SWITZERLAND Ltd. | | REV. 3 | | B767-300 | Cabin reconfiguration - Air Seychelles 767-300 Aircraft reconfiguration by replacing the existing J-class seat with an electrically powered seat. There are adaptations to the existing PSU, Y-class cabins to suit new layout. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate. Certification basis as per TCDS plus: - CRI H-01/Special Condition H-01 "Enhanced Airworthiness Pr (...) | Change Classification and Certification Sheet doc. Ref. C767-25-1010 Rev 03 dated 18 November 2010; Master Data List doc. Ref. C767-25-1010-MDL Rev 03 dated 18 November 2010; Instructions for Continued Airworthiness doc. Ref. C767-25-1010-ICA-01 Rev (...) | None. | 26.11.2012 |
| 10034187 | SR TECHNICS SWITZERLAND Ltd. | | | | A330-300 | Virgin Atlantic Airways (VAA) A330-300 EIS Cabin Reconfiguration (59W /255Y). This STC reconfigures the cabin layout from 48 W class seat / 265 Y class seat to 59 W class seat / 255 Y class seat, with associated adaptations. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI D-3323-001; Special Condition applicable to the installation of the AmSafe Aviation (...) | SR Technics Master Document List, ref : S.OSMD.000001-MDL, dated 14 March 2011 or later revisions of the above listed documents approved by EASA | 1-, This STC must be implemented together with EASA STC ref: 10031808 Rev 01 (and associated Approved Model List), or later revisions approved by EASA. 2-, This STC is applicable to A330-300 following MSN : -, 1195 -, 1201-, 1206-, 1215-, 1231 | 14.03.2011 |
| 10038294 | SR TECHNICS SWITZERLAND Ltd. | | | | A310 | Amiri Flight A310-300 Cabin Refurbishment by refurbish the existing Cabin by: - Removal of the existing seats and installation of new seats in the First, Business and Economy class. - Replace the existing IFE system- Installation of a new VCC, class divid (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. SCS or ESFS; Voluntary compliance for FAR 25.853(b) to FAR 25 Amdt. 59 Derogation to lat (...) | Master Document List, Doc. No. S.OSMV.000003-MDL, Rev. 00- AFMS, Doc. No. S.OSMV.000003-AFMS, Rev. 00- ICA, Doc. No. S.OSMV.000003-ICA, Rev. 00 or later revisions of the above listed documents approved by EASA | S/N: 0667 A7-AFE | 03.02.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|--------------|----------------|----------------------|---|---|--|---|--|---------------|
| 10040981 | SR TECHNICS SWITZERLAND Ltd. | | | | A340-541 | VIP Cabin RefurbishmentThe modification consist of the following items:- „Refurbishment of existing cabin materials- „Deletion of a double seat and installation of 8th stowage- „Installation of a book cabinet- „Upgrade of the IFE equipment(monitors, (...) | JAR 25, Amdt. 14The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approval.SCs or ESFS:- „ORI A-1301-003, Existing Exemption & Deviation, ESF/De (...) | - Master Document List, Doc. No. S.OA.EV.000008-MDL, Rev. 01-AFMS, Doc. No. S.OA.EV.000008-AFM, Rev. 00- ICA, Doc. No. S.OA.EV./000008-ICA, Rev. 00or later revisions of the above listeddocuments approved by EASA | S/N: 0495 A7-HHH | 10.08.2012 |
| 10042831 | SR TECHNICS SWITZERLAND Ltd. | | REV. 1 | | A330-223 | Cabin reconfigurationBy changing the existing Air Berlin Business Class InteriorConfiguration into a new Ethad Business Class Interior Configuration. | JAR 25 Amdt: 13The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | - Master Data List, Doc. No. S.OA.EZ.000180-MDL, Rev. 02, dated03.05.2013- ICA, Doc. No. S OAEZ 000180-ICA, Rev. 00, dated 06.11.2012- ICA, Doc. No. C OAEZ 000516-ICA<->- Rev. 00, dated 02.04.2013or later revisions of the above listed documents approved by EASA. | S/N: 0403 / 0432 / 0444 / 0454 / 0469 / 0476 / 0493 / 0739 / 0828 /0911.Approval is valid only for MSN mentioned above.Prior to installation of this modification it must be determined thatthe interrelationship between this modification andany other (...) | 24.05.2013 |
| 10044593 | SR TECHNICS SWITZERLAND Ltd. | | REV. 1 | | A340-313 | THT A340 Cabin & IFE RetrofitThe change consists out of:Cabin layout- Removal of all passenger seating and installation of new Business andEconomy seats.- Removal of some monuments and installation of new build galleys, stowages & Partitions.- Rewor (...) | The Certification Basis for the original product as amendedby thefollowing additional or alternative airworthiness requirements: SCD: 1322-002The certificated noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | - Master Document List, Doc. No. S.OA.ED.000046-MDL, Rev. 01, dated15.05.2013- AFMS, Doc. No. S.OA.ED.000046-AFM01, Rev. 00, dated 25.03.2013- AFMS, Doc. No. S.OA.ED.000046-AFM02, Rev. 00, dated 28.03.2013- ICA, Doc. No. S.OA.ED.000046-ICA, Rev. 01, dat (...) | S/N: 0385 / 0438 / 0446 „F-OJGF / OSEA / OSUNPrior to installation of this design change it must be determined thattheinterrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noadverse effect upon the airworthiness of the product. | 24.05.2013 |
| 10038822 | ST AEROSPACE MOBILE, INC. | | | FAA STC ST13005AT-T | B757-200 | Conversion of Passenger Configuration to Cargo/Passenger CombinationConfiguration | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate.For Significant Changed Areas, Certification Basis is CS25 Amendment 9including the additional:S (...) | - ST Aerospace Mobile Inc., MDL: 33600000, Rev B, 21/12/2011-AMM Supplement: 757SC AMM-01, Rev 1, 07/12/2011-MPD Supplement: 757SC-MPD-01, Rev 7, 02/03/2012-AFM supplement: 757SC-AFM-01, Rev 1, 05/03/2012-ICA Supplement: 336A/ICA/001, Revision E, 01/ (...) | This STC is limited to a/c serial number 25622 only/As per the Temporary Deviation a/c converted under this STC cannot beoperated after 23/12/2013 unless incorporating design changes to theMain Deck Cargo Door.This STC cannot be installed on aeroplan (...) | 22.03.2012 |
| 10027268 | STANDARD AERO | | | FAA STC ST02756CH-D | 737-600, -700, -800, -900, 737-900ER | Installation of an in flight entertainment (IFE) system | In accordance with TCDSThe Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval | Standard Aero Documents:Master Data List MDL1104-100 Rev. Bdated September 09, 2009or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | see limitations and conditions on STC FAA ST02756CH-D | 17.09.2009 |
| 10028167 | STANDARD AERO | | | FAA STC ST01991CH | MF50 | EASA Validation of FAA STC ST01991CHInstallation of an L3 Communications (Fairchild) Model F1000 Flight DataRecorder (FDR) System in accordance with Garrett Aviation ServicesDrawing List, Doc. Number 1002382 Revision H, dated 04th June 2004 or later (...) | The Certification Basis for the original product remains applicable to this certificate/ approvalThe certificated noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval | -Garrett Aviation Services Drawing List, Doc. Number 1002382 Revision H, dated 04th June 2004- Aeroplane Flight Manual Supplement, Garrett Aviation Services Doc. No.1002752, Revision A, originally approved 03rd June 2004-Airworthiness Limitations Garr (...) | None | 07.12.2009 |
| 10029879 | STANDARD AERO | | | FAA STC SE02357NY | PT6A-135A | Installation of the Pratt & Whitney Canada, PT6A-135, dual fuel feedline configuration in accordance with Standard Aero, Inc. EngineeringReport, 001-07-01, Rev. 0, dated February 07, 2007. | FAR Part 33, effective 1 February 1965, and Amendments 33-1 to 33-InclusiveThe Certification Basis for the original product remains applicable to this certificate/ approval | Standard Aero, Inc. Instructions for Continued Airworthiness, ReportNumber 001-07-ICA, Rev. 0., dated February 07, 2007, or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | Not Applicable | 28.04.2010 |
| 10032484 | STANDARD AERO | | REV. 1 | FAA STC SA02367NY | F90 KING AIR | Installation of Pratt and Whitney Canada PT6A-135A Engines. | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noiseand/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | -Modified with dual fuel feed lines, in accordance with Standard AeroInc, Installation Instructions, Report, 55055-8, Rev 0 dated 26 February 2007.-PT6A-135A engines must be modified in accordance with Supplemental TypeCertificate SE02357Nyprior to i (...) | None | 23.11.2010 |
| 10035155 | STANDARD AERO | | | FAA STC ST02803CH-D | EMB-135ER, EMB-135LR, EMB-135KE, EMB-135KL, EMB-145, EMB-145MP, EMB-145EP, EMB-145ER, EMB-145MR, EMB-145LR, EMB-145XR | Installation of a Honeywell KHF-1050 HF System. | The Certification Basis for the original product remains applicable to this certificate/ approval.EASA TCDS IM.A.032The certificated noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | - Master Data List : 1018147 Rev.A dated 05 February 2010- Airplane Flight Manual Supplement : 1018197 Rev.A dated 21 February2010- Instruction for Continued Airworthiness : 1018140 Rev.A dated 13January 2010- EWIS ICA : 1019049 Rev.B dated 23 July (...) | None. | 30.05.2011 |
| 10038105 | STANDARD AERO | | | FAA STC ST02912CH-D | MF50 | Installation of an interior refurbishment and a pocket door/Validation of FAA STC N° ST02912CH-D | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Standard Aero Master Data List doc. N° 1019814 Revision B dated January08, 2011.Instructions for Continued Airworthiness doc. N° 1019827 Revision Bdated January 05, 2011.FAA STC N° ST02912CH-D issue dated January 09, 2011or later revisions of the abo (...) | None | 19.01.2012 |
| 10040181 | STANDARD AERO | | | FAA STC SA02715CH-D | 1. 200 2. 200C 2. B200 2. B200C | Installation of Pratt & Whitney PT6A-52 Engines in accordance withStandard Aero Master Data List document number 1016151, Revision "A",dated June 01, 2009 or later FAA approved revision. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approval/as established by FAA STC SA02715CH-DThis certificate/ approval involves a change to i (...) | Airplane Flight Manual Supplement document number 1016494 Revision "A"dated June 08, 2009<->->-Instructions for Continued Airworthiness document number 1016175,Revision "A"<->->-TCDSN EASA.IM.A.277 records C10727 through C10730<->->-or later revi (...) | None. | 15.06.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|------------------------|----------------|----------------------|--|---|---|--|--|---------------|
| 10041237 | STANDARD AERO | | REV. 2 | FAA STC ST03038CH-D | GV | Reconfiguration of a Cabin Floorplan Arrangement | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -FAA STC ST03038CH-D issued 19th December 2011-Installation in accordance with Standard Aero Master Data List - MDL-1021004, Revision H, dated 19th December 2011 for later revisions of the above listed documents approved by EASA in accordance with EASA E (...) | 1. The installer must determine whether this design change is compatible with previously approved modifications. 2. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written (...) | 05.09.2012 |
| 10043464 | STANDARD AERO | | | FAA STC ST03033CH-D | G-1159A, G-IV, GV, GV-SP, GIV-X | Installation of a True North Simphone System with Global Swift Broadband. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -, Standard Aero Master Data List, ref. doc. 1021093, Revision D, dated 16th December 2011, -, FAA Approved Airplane Flight Manual Supplement, ref. doc. 1021218, Revision A, dated 16th December 2011, -, Airworthiness Limitations specified in Standard Aer (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 30.01.2013 |
| 10043526 | STANDARD AERO | | | FAA STC ST03175CH-D | A330-301, A330-302, A330-303, A330-322, A330-323, A330-341, A330-342, A330-343 | Installation of the ARINC Cabin Connect provisions. Installation of the ARINC Cabin Connect in-flight entertainment system provisions in preparation for activation on-board Airbus A330-300 series aircraft. The connection or activation of the components (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List DC104-9157-00, dated 01/17/2013, FAA Approval ST03175 CH-D, dated 30/01/2013, Statement of Compliance, later 20130201 Foreign Valid 3175 EASA, received 01/02/2013 or later revisions of the above listed documents approved by EASA in (...) | The connection or activation of the components installed by this modification is not covered by this STC. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed (...) | 04.02.2013 |
| 10045665 | STANDARD AERO | | | FAA STC ST03225CH | ERJ 170-100 LR ERJ 170-100 STD ERJ 170-200 LR ERJ 170-200 STD | Installation of a portable emergency locator transmitter (ELT) and modification of interior placards. EASA validation of Basic FAA STC ST03225CH. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Standard Aero Master Data List 1024554, Revision (C), dated June 25, 2013 or later revisions of the above listed documents approved by EASA. | None | 10.07.2013 |
| 10043973 | STANDARD AERO (ALLIANCE), INC. | | | FAA STC ST03176CH-D | A330-301/-302/-303/-321/-322 A330-323/-341/-342/-343 | Activation of the ARINC Cabin Connect System | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. | -FAA Approval ST03176CH-D, dated 14/02/2013-Instructions for Continued Airworthiness DC113-9268-00, Rev. A, dated 07/02/2013-Aircraft Flight Manual Supplement DC303-9268-00, Initial Release, dated 14/02/2013-Master Data List DC104-9268-00, Rev. D, da (...) | Before the installation of this STC shall be applied the following STCs: 10043526 provisions for Cabin ARINC Connect System-10043936 Compatibility with wireless and cellular data technologies Prior to installation of this design change it must be det (...) | 08.03.2013 |
| 10015708 | STAR AVIATION Inc. | EASA, IM, A.S. 01701 | REV 1 | FAA STC ST02943AT | BOEING 777-200 BOEING 777-300 | Installation of Connexion by Boeing Next Generation Antenna and Satellite Data System. Note: This Revision reflects the transfer of this STC from Flight Structures, Inc. to Star Aviation, Inc. | | MDL R44-0211, Revision AM, dated 20th June 2006 or later FAA approved revision, -AFM Supplements R44-0397, Revision B, dated 20th June 2006 for the -200 series or R44-0220, revision B, dated 20th June 2006 for the -300 series, or later FAA approved Revision | 11. Airworthiness Limitations Section of the Instruction for Continued Airworthiness, document numbers R44-0403, revision B, dated 22nd July 2005 for -200 series, R44-0218, revision B, dated 20th October 2004 for -300 series, or later FAA approved (...) | 05.09.2007 |
| 10015850 | STAR AVIATION Inc. | EASA, IM, A.S. 01950 | | ST02989AT | CESSNA 500 CESSNA 501 CESSNA 550 CESSNA 551 CESSNA S550 | The modification will consist of a composite structure with metal faceplate which will be fitted over the center of the existing glare shield. An area of the existing glare shield will be removed to allow for wires to be routed to the annunciators and (...) | | Master Data List R25-0283 Rev. E, dated 17 January 2006 AFM Supplement Doc. Nr. R25-342, Rev. IR dated 24 November 2004 | | 12.05.2006 |
| 10036456 | STAR AVIATION Inc. | P-EASA, IM, A.S. 00285 | REV. 1 | FAA STC ST02861AT | B747-400 | Installation of Connexion by Boeing Next Generation Antenna and Satellite Data System in accordance with Star Aviation Inc. Master Data List R44-0163, Revision CP, Dated May 12, 2011 or later approved revision | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The certification basis for parts changed or affected by the change since thereference (...) | Aircraft Flight Manual Supplement R44-0300 Rev. IR dated October 16, 2004 Instruction for Continued Airworthiness Star Aviation Document NoR44-0273, Rev. D, dated June 20, 2011 for later revisions of the above listed documents approved by EASA in accordance (...) | None | 09.09.2011 |
| 10041648 | STAR AVIATION Inc. | EASA, IM, A.S. 00286 | REV. 1 | FAA STC ST02942AT | 747-400 | Installation of Cabin Wireless Local Area Network (CWLAN) in accordance with Star Aviation Inc. Master Data List R44-0175, Revision G, dated October 13, 2004, or later FAA approved revisions and Provisional installation of a Panasonic Avionics Corporation (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval 14 CFR Part 25 Amendment 25-0 through 123 for the following requirements 25.23, 25.29, (...) | Master Data List R44-0175, Revision G, dated October 13, 2004 (->-) Master Data List R44-0613, Revision D, dated July 19, 2012 (->-) AFM Supplement R44-0293, Revision A, dated October 18, 2000 Instructions for Continued Airworthiness Document R4 (...) | None | 23.10.2012 |
| 10016830 | STARFLEX L.L.C. | EASA, IM, R.S. 01201 | | FAA STC SH7962SW-D | EUROCOPTER FRANCE AS 355F2 | Installation of two Allison 250-C20R engines and enhanced anti-torque system in accordance with AHC drawing list 355A82-1755 Revision D dated November 5, 1990 or later FAA approved revision | | Instructions for Continued Airworthiness: Starflex LLC document reference C20RF2MMS, edition 1 amendment 0 dated 11 January 2007, or later revision. Operation: AHC Rotorcraft Flight Manual Supplement reference 355-FMS-011, dated 01 February 1991 approved by FAA, or later FAA approved revision. | 1. The present STC is the validation of the FAA STC SH7962SW-D, last amended on 01 February 2006. 2. Compatibility of this modification with previously approved modifications of the aircraft must be determined by the installer. | 16.01.2007 |
| 10016832 | STARFLEX L.L.C. | EASA, IM, R.S. 01203 | | SH7961SW-D | | Installation of two Allison 250-C20R engines in accordance with AHC drawing list 355A82-1753 Revision H dated November 5, 1990 or later FAA approved revision | | Instructions for Continued Airworthiness: Starflex LLC document reference C20RF1MMS, edition 1 amendment 0 dated 11 January 2007, or later revision. Operation: AHC Rotorcraft Flight Manual Supplement reference 355-FMS-010, dated 01 February 1991 approved by FAA, or later FAA approved revision. | 1. The present STC is the validation of the FAA STC SH7961SW-D, last amended on 01 February 2006. 2. Compatibility of this modification with previously approved modifications of the aircraft must be determined by the installer. | 16.01.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|----------------|----------------|----------------------|--|---|---|--|--|---------------|
| 10035081 | STC ACQUISITION, LLC | | | FAA STC ST01567LA | G-1159 (GULFSTREAM II) G-1159A (GULFSTREAM III) G-1159B (GULFSTREAM IIB) | Installation of a pylon mounted ejector system with new design thrustreversers, to show compliance with ICAO Annex 16, Volume 1, Amendment 9, Chapter 3 noise requirements. | The Certification Basis for the original product remains applicable to this certificate/ approval. CAR 4b and FAR 25 and FAA Issue Paper P-1 as defined in FAA STC ST01567LA. This certificate/ approval involves a change to the requirements forenvironment (...) | Installation in accordance with Master Drawing List ref A-<(>&-<)>C-00-00-051 Rev N, dated 17 March, 2005. Operation in accordance with Aircraft Flight Manual Supplement ref A-<(>&-<)>C-01-00-270, dated 16 May, 2005 for model G-1159 (GulfstreamII)Operati (...) | Model G-1159, (Gulfstream II)1), Aircraft without Aircraft Service Change (ASC) 81, #62,500 Pound increased Gross Weight Modifications, or those with a Maximum (ramp) weight less than 62,500 lb are not eligible for this modification.2), Aircraft with A (...) | 24.05.2011 |
| 10014128 | STC TWENTY ONE LIMITED | EASA.A.S.01236 | REV 1 | | DC-9-80 (MD-80) SERIES | Introduction of FlightVu Cabin Surveillance System | | Aero Engineering Design Modification Certificate 25-77-433 dated April 12,2005 or later approved revision | | 21.06.2006 |
| 10014129 | STC TWENTY ONE LIMITED | EASA.A.S.01239 | | | BOEING 757-200 | Introduction of SSR Mode S Enhanced Surveillance | | -Aero Engineering Design Modification Certificate 34-15-233-Aero Engineering Design AFM Supplement AED/0757/004 Issue 1 or later approved revision | | 30.05.2005 |
| 10014131 | STC TWENTY ONE LIMITED | EASA.A.S.01241 | | | BAE ATP | Rework to Landing Gear Emergency Up-Lock Release Mechanism | | STC 21 Master Document List S21.MDL-0013STC21 AFM Supplement S21.FMS-ATP-006 | | 15.11.2005 |
| 10014185 | STC TWENTY ONE LIMITED | EASA.A.S.01323 | REV 1 | | FOKKER F.27 MARK 500 FOKKER F.27 MARK 600 | Use of TRA-45A Display Indicator | | - STC Twenty One Modification S21.34-62-1013- STC Twenty One AFM Supplement AED/F27/8 Issue 3 or later approved revision | | 30.01.2006 |
| 10014209 | STC TWENTY ONE LIMITED | EASA.A.S.01361 | REV 1 | | BAE ATP | Introduction of Class E Compartment for LFD Modified Aircraft. | | STC Twenty One Limited Master Document List S21.MDL-0017, initial issue dated 4th April 2008 or later approved revisions. | 1. Flight Manual Supplement S21.FMS-0008; | 09.04.2008 |
| 10014210 | STC TWENTY ONE LIMITED | EASA.A.S.01362 | REV 1 | | BAE ATP | Introduction of Freight Loading & Restraint System for LFD Modified Aircraft. | | STC 21 Master Drawing List No. S21.MDL-0018 initial Issue dated 25 January 2008 or later approved revision. STC 21 Airplane Flight Manual Supplement No. S21-FMS-0009 at Initial Issue or later approved revision. | This major change is limited to embodied into a Bae aircraft previously modified for the installation of a large cargo door at the rear of the fuselage in accordance with Bae Service Bulletin ATP-52-036 and with the installation of a main cabin Class (...) | 09.04.2008 |
| 10014211 | STC TWENTY ONE LIMITED | EASA.A.S.01363 | | | BAE ATP | Installation of AAR Freight Loading System | | -STC21 Change No S21.25-23-1005-STC21 AFM Supplement SerialNo S21.FMS-atp-0010 Issue No. or later approved revision | This STC may only be embodied into Bae ATP aircraft previously modified for the installation of a large cargo door at the rear of the fuselage, in accordance with British Aerospace Service Bulletin SBATP-52-036 (Modification M60138A), with a barrier (...) | 30.01.2006 |
| 10014252 | STC TWENTY ONE LIMITED | EASA.A.S.01477 | | | BOEING 757-200 | Introduction of Honeywell MKV Enhanced Ground Proximity Warning System (EGPWS) | | Master Data List S21.MDL-0001-Issue 2 or later approved revision. AFM Supplement S21.FMS-0001 Issue 1 or later approved revision. | | 22.09.2005 |
| 10014253 | STC TWENTY ONE LIMITED | EASA.A.S.01478 | | | BOEING 757-200 | Introduction of Honeywell GNSSU Receiver | | Master Data List S21.MDL-0002-Issue 1 or later approved revision. | | 22.09.2005 |
| 10014344 | STC TWENTY ONE LIMITED | EASA.A.S.01606 | | | LOCKHEED L1011 - TRISTAR | Introduction of a stand-alone GPS receiver | | STC Twenty One AFM Supplement S21.FMS-0007 Issue 1 or later approved revision* | | 22.03.2006 |
| 10014392 | STC TWENTY ONE LIMITED | EASA.A.S.01681 | | | SAAB SF340A SAAB SF340B | Saab 340 Class E Cargo Compartment This significant major change, designed by STC WP-EASA.A.S.01681 Twenty One Limited, will convert the main passenger cabin to a class E cargo compartment for the carriage of freight | | Airplane Flight Manual Supplement, Master Minimum Equipment List Supplement, Mass and Balance Report, Maintenance Programme, Maintenance Manual, Wiring/ Schematic Manual, Illustrated Parts Catalogue | -- | 12.02.2007 |
| 10014417 | STC TWENTY ONE LIMITED | EASA.A.S.01738 | | | SAAB 340 SAAB 340B | Saab 340 Bulk Cargo Restraint System (as defined in document S21.MDL-0034). This major change will introduce transverse nets at discrete positions in the aircraft main cabin (converted to Class E under STC Twenty One Limited major change S21.25-63-10 (...) | | S21.FMS-0014 Flight Manual Supplement S21.TEC-0028 Weight and Balance Manual S21.TEC-0116 Supplement Manual for Instruction of a Bulk Cargo Restraint System Maintenance Programme Maintenance Manual Illustrated Parts Catalogue | This STC must be used in conjunction with EASA STC EASA.A.S.01681 "Saab 340 Class E Cargo Compartment". | 23.04.2007 |
| 10014418 | STC TWENTY ONE LIMITED | EASA.A.S.01739 | | | SHORTS SD3-60 | Introduction of ACAS II & Single Mode S Transponders with Elementary Surveillance. | | STC Twenty One Modification S21.34-39-1014 AFM Supplement S21.FMS-0015 | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 13.03.2006 |
| 10014443 | STC TWENTY ONE LIMITED | EASA.A.S.02494 | | | BAE ATP | Introduction of Rockwell Collins ACAS II and Dual Mode S Transponder System | | STC21 Change S21.34-23-1008 STC21 AFM Supplement S21.FMS-ATP-0002 Issue 1 or later approved revision | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 16.02.2006 |
| 10014446 | STC TWENTY ONE LIMITED | EASA.A.S.02499 | | | CESSNA 550 CITATION II | Introduction of an AirCell Satellite Telephone System ST 3100 to the aircraft. This modification installs the handset into an arm rest on the rear starboard side of the aircraft. The Satellite Phone Transceiver is installed in the mid cabin drinks sto (...) | | Installation: Accomplishment Instruction, Serial No.: S21.ACC-0026, Iss. 1 Aircraft Maintenance Manual, Wiring Diagram Manual, Illustrated Parts Catalogue, refer to STC Twenty One Ltd Supplemental Manual S21.TEC-0032, Drawing No. Title 11-29-10004 (...) | This design change should be embodied concurrently with or after the embodiment of STC Twenty One Limited minor design change S21.23-29-1000. | 09.06.2006 |
| 10014447 | STC TWENTY ONE LIMITED | EASA.A.S.02500 | | | BOEING 767-200-300 | Introduction of SSR Mode S Enhanced Surveillance System. | | Master Document List S21.MDL-0049 AFM supplement S21.FMS-0028 Issue 1 | Boeing 767-200/300 series. It is a condition of this STC that transponders of different manufacturers must not be intermixed, refer to STC 21 IPC supplement S21.TEC-0033 for further details. | 12.10.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|----------------|----------------|----------------------|--|---|---------------------|--|--|---------------|
| 10014477 | STC TWENTY ONE LIMITED | EASA.A.S.02539 | | | BAE 146 | Introduction of T2 CAS Collision Avoidance & Terrain Awareness System | | STC Twenty One Modification S21.34-25-1017STC Twenty One AFM Supplement S21.FMS-0016 Issue 1 or later approved revision | | 22.05.2006 |
| 10014522 | STC TWENTY ONE LIMITED | EASA.A.S.02605 | | | AIRBUS A310 SERIES | Installation of Honeywell MK V EGPWS. This supplemental TypeCertificate installs a Terrain Awareness Warning System. | | Master Document List, ref. S21.MDL-109 Issue 1, dated 31 August 2006 (or later approved revisions of this document are acceptable). Aircraft Flight Manual Supplement, ref. S21.FMS-0017 Issue 1, dated 12 November 2007 | | 28.11.2007 |
| 10014523 | STC TWENTY ONE LIMITED | EASA.A.S.02606 | | | BOEING 727 SERIES | Introduction of Emergency Locator Transmitter | | STC21 Modification S21.25-12-1055 | | 15.06.2006 |
| 10014564 | STC TWENTY ONE LIMITED | EASA.A.S.02664 | | | BOMBARDIER CL-600-2B16 (604) | Introduction of Passenger Cabin Interior | | STC Twenty One MDL No. S21.MDL-0131 at Issue 1 or later approved revision. Midcoast Aviation Inc. AFM supplement 34-261M021 or later approved revision | | 22.06.2006 |
| 10014597 | STC TWENTY ONE LIMITED | EASA.A.S.02717 | | | BAE ATP | Rework of LH Passenger Door | | STC Twenty One Master Drawing List S21.MDL.0135 Initial Issue dated 16 April 2007 or later approved revision. STC TwentyOne AFM Supplement S21.FMS-0018 Issue 1 or later approved revision. | | 11.05.2007 |
| 10014628 | STC TWENTY ONE LIMITED | EASA.A.S.02766 | | | BAE 146/AVRO 146 SERIES | Introduction of T2CAS Collision Avoidance & Terrain Awareness Warning System | | STC Twenty One Modification S21.34-86-1041STC Twenty One AFM Supplement S21.FMS-0020 Issue 1 dated 18 October 2006 or alter approved revision | -- | 03.11.2006 |
| 10014667 | STC TWENTY ONE LIMITED | EASA.A.S.02824 | | | BOEING 757-200 SERIES | Reduction of Over-Wing Emergency Exits from 4 to 2. | | AFM Supplement No. S21.FMS.0021. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 22.12.2006 |
| 10014683 | STC TWENTY ONE LIMITED | EASA.A.S.02846 | | | SD3-60 | Introduction of Honeywell MK VI EGPWS | | Modification S21.34-39-1047Flight Manual Supplement S21.FMS+-0022 Issue 1 | | 07.12.2006 |
| 10014711 | STC TWENTY ONE LIMITED | EASA.A.S.02883 | | | BAE 146 | Introduction of Hoenywell RDR-4B Weather Radar System | | STC Twenty One Modification S21.34-25-1049STC Twenty One Master Document List S21.MDL-0240 Issue 1 | | 10.07.2007 |
| 10014723 | STC TWENTY ONE LIMITED | EASA.A.S.02900 | | | FOKKER F.27 MARK 050 FOKKER F.27 MARK 0502 FOKKER F.27 MARK 0604 FOKKER F.27 MARK 100 FOKKER F.27 MARK 200 FOKKER F.27 MARK 300 FOKKER F.27 MARK 400 FOKKER F.27 MARK 500 FOKKER F.27 MARK 600 FOKKER F.27 MARK 700 | Introduction of Rockwell Collins Dual ADF-60 System in the F27 Mark 200. This installation replaces the existing BendixDual ADF System. In accordance with STC Twenty One Limited Master Drawing List document S21.MDL-0264 dated 14/12/2006 or later EASA approved revision. | | Data Package containing the following documents, and additional supporting data: STC Twenty One Limited Master Drawing List document S21.MDL-0264. Certification Programme STC TwentyOne Limited document S21.PROG-0265. Compliance Record STC Twenty One Li (...) | Instructions for Continued Airworthiness: STC Twenty One Limited Change Supplement Manual document S21.TEC-0094 Revision 0, dated 21/03/2002, or later EASA approved revision. | 20.06.2007 |
| 10014814 | STC TWENTY ONE LIMITED | EASA.A.S.03022 | REV 2 | | BOMBARDIER CL-600-2B19 | Introduction of Passenger Cabin interior. | | 1. STC 21 Master Document List No. 21.MDL-0316 Issue 1 dated 10 January 2008 or later approved revision. 2. STC 21 Flight Manual Supplement No. S21.FMS-0033 Issue 1 or later approved revision. | EASA STC No. EASA.IM.A.S.01752 must be fitted. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect (...) | 16.01.2009 |
| 10014900 | STC TWENTY ONE LIMITED | EASA.A.S.03130 | | | 737- 600, 700, 800, 900 737-200, 300, 400, 500, | Installation of Cabin Vu-123 Cabin Surveillance System. | | MDL S21. MDL-0351 Issue 1, dated 21st January 2008 or later EASA approved revisions. | | 23.01.2008 |
| 10014918 | STC TWENTY ONE LIMITED | EASA.A.S.03164 | | | BOMBARDIER CL-600-2B16 (604) | Introduction of Passenger Cabin Interior. | | STC 21 Master Document List No. S21.MDL-0350 at Issue 1 dated 19 August 2008 or later approved revision. STC 21 Flight Manual Supplement No. S21. FMS-0041 at Issue 1 or later approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 25.08.2008 |
| 10014997 | STC TWENTY ONE LIMITED | EASA.A.S.03297 | REV 1 | | BOEING 767-200 BOEING 767-300 | Introduction of CabinVu-123 Cabin Surveillance System | | STC 21 MDL S21.MDL-0393, Issue 1, 20/12/2007 | | 20.05.2008 |
| 10015018 | STC TWENTY ONE LIMITED | EASA.A.S.03324 | | | CESSNA 500, CITATION CESSNA 550 | Introduction of Dual GNS-430/530 | | STC Twenty One Ltd Major Change S21.34-29-1085Flight Manual Supplement S21.FMS-0032 Issue 1 | | 05.05.2008 |
| 10015068 | STC TWENTY ONE LIMITED | EASA.A.S.03394 | | | AIRBUS A318 AIRBUS A319 AIRBUS A320 AIRBUS A321 | Introduction of Wingspeed XLLink Communications and Data Management System | | 1. STC Twenty One Limited Major Change S21.23-02-10392. Flight Manual Supplement S21.FMS-0042 Issue 1 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 29.10.2008 |
| 10015109 | STC TWENTY ONE LIMITED | EASA.A.S.03452 | | | BOMBARDIER CL-600-2B19 | Introduction of Passenger Cabin interior. | | STC 21 Master Document List No. S21.MDL-0482 Issue 1 dated 3 March 2008 or later approved revisions. STC 21 Flight Manual Supplement No. S21.FMS-0034 at Issue 1 or later approved revision. | | 07.03.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|------------------|----------------|----------------------|--|--|---|--|--|---------------|
| 10015115 | STC TWENTY ONE LIMITED | EASA.A.S.03464 | | | SAAB SF340A | Rework of Class E cargo role of QC variant | | S21.FMS-0038 Flight Manual Supplement S21.MMEL-0011 Master Minimum Equipment List Supplement S21.TEC-0105 Supplement Manual for Continued Airworthiness S21.TEC-0206 Weight and Balance Manual | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 19.12.2008 |
| 10015176 | STC TWENTY ONE LIMITED | EASA.A.S.03553 | | | BOEING 737-300 | Carriage of Horses & Grooms | | 2 Master Document List S21.MDL-0524 Issue 5* 2 Airplane Flight Manual Supplement S21.FMS-0047 Issue 1* 2 Instructions for Continued Airworthiness S21.TEC-192 Issue 1** or later approved revisions of those documents. | 1. This design change is applicable to Boeing 737-300 aircraft converted under FAA STC SA2968SO, SA2969SO, and SA2970SO & SA2971SO as previously been validated by the UK CAA under Airworthiness Approval Note AAN 24984.2. The certification basis (...) | 07.04.2009 |
| 10015185 | STC TWENTY ONE LIMITED | EASA.A.S.03568 | | | BOMBARDIER BD-700-1A11 | Introduction of Passenger Cabin Interior. | | - STC Twenty One Ltd Master Document List S21.MDL-0532 dated 09 December 2008 or later approved revision.- STC Twenty One Ltd Flight Manual Supplement S21.FMS-0048 Issue 1 or later approved revision. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 23.12.2008 |
| 10015199 | STC TWENTY ONE LIMITED | EASA.A.S.03590 | | | BOEING 767-200 | Introduction of Change 7.0 ACAS II and Dual Mode S | | STC 21 Master Data List S21.MDL-0543 Initial issue, 12/06/2008 | | 12.06.2008 |
| 10017185 | STC TWENTY ONE LIMITED | EASA.R.S.01318 | | | AS 355 F1 | Introduction of Skylark 2 Radar. | | Modification - S21.25-10-1115.Maintenance Manual - S21.TEC-0104.AFM Supplement - S21.FMS-0024. | | 01.02.2007 |
| 10017212 | STC TWENTY ONE LIMITED | EASA.R.S.01368 | | | R44 R44 II | Installation of removable crop spraying equipment onto Robinson R 44 & R 44 II helicopters | | - STC Twenty One Limited Rotorcraft Flight Manual Supplement S 21.FMS-0029 dtd. 27 July 2007 or later approved revisions- STC Twenty One Limited Major Change Design Change S21.25-74-1172 - STC Twenty One Limited Master Document List S21.MDL- (...) | 1. Certified in Restricted Category.2. All limitation of RFM supplement apply.3. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed m (...) | 01.08.2007 |
| 10026853 | STC TWENTY ONE LIMITED | P-EASA.A.S.03708 | REV. 1 | | BOMBARDIER CL-600-2B19 | Introduction of Wireless Local Area Network. | EASA.IM.023 plus CRI F-01. | - S21.25-89-1302 - MDL-0580 Issue 4. | | 20.08.2009 |
| 10028521 | STC TWENTY ONE LIMITED | | | | BD-700-1A10 | Customized Aircraft Completion - ASN 9251. | EASA.IM.A.009.The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.CRI F-01 "Installation of Wireless System".The certificated noise and/ or (...) | STC Twenty One Ltd Master Document List No. S21.MDL-0726 dated 15 January 2010.STC Twenty One Ltd Airplane Flight Manual Supplement No. S21.FMS-0057.Instructions for Continued Airworthiness Midcoast Aviation Inc. Supplemental Maintenance Manual No.35-0 (...) | None. | 19.01.2010 |
| 10029416 | STC TWENTY ONE LIMITED | P-EASA.A.S.03895 | | | BD-700-1A10 | Customized Aircraft Completion - ASN 9266 | EASA.IM.A.009The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.CRI F-01 "Installation of Wireless System".The certificated noise and/ or (...) | STC Twenty One Ltd Master Document List No. S21.MDL-0655.STC Twenty One Ltd Airplane Flight Manual Supplement No. S21.FMS-0054.Instructions for Continued Airworthiness Midcoast Aviation Inc. Supplemental Maintenance Manual No. 35-027E134, or later rev (...) | None. | 24.03.2010 |
| 10030161 | STC TWENTY ONE LIMITED | | | | 1. CESSNA 337 2. REIMS CESSNA 337 | Introduction of Bubble Windows | The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | STC Twenty One Limited Master Document List No. S21.MDL-0752 at Initial issue dated 26 May 2010 or later revisions of the above listed documents approved by EASA | Restricted to un-pressurized versions of the Cessna/Cessna Reims 337 aircraft | 31.05.2010 |
| 10030229 | STC TWENTY ONE LIMITED | P-EASA.A.S.03972 | | | B757-200/-300 | Introduction of Air Conditioning Filter | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.As per FAA TCDS A2NM and as modified by CRI D-01 and CRI H-01The requirements for environmental protection (...) | STC 21 Compliance Record, S21.COMP-0694, Issue 1, 03/06/2010STC 21 Master Document List, S21.MDL-0694, Issue 1, 03/06/2010 or later revisions of the above listed documents approved by EASA | Installation of STC21 Limited Minor Change S21.21-15-1004, must be embodied prior to (>-<)-> or concurrent with (>-<)-> this STC | 04.06.2010 |
| 10030237 | STC TWENTY ONE LIMITED | P-EASA.A.S.03926 | | | SAAB SF340A, SAAB 340B | Rework of the Class E Compartment for inclusion of a courier seating area for up to 5 occupants, S21.25-63-1355. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.2The requirements for environmental protection and (...) | S21.MDL-0663 Issue 1 Master Document List, dated 28 May 2010 including reference to AFM Supplement, MMEL, Maintenance Manual including ICA and EWS ICA, W (-&-&-)B Manual, etc. or later revisions of the above listed documents approved by EASA | EASA.A.S.01681 R1 and EASA.A.S.01738 must be installed | 04.06.2010 |
| 10030991 | STC TWENTY ONE LIMITED | | REV. 1 | | L-410 UVP-E20 CARGO L-410 UVP-E9 L-410 UVP, L-410 UVP-E. | Introduction of under wing emergency exits. | CS-23 Amendment 11The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are (...) | Compliance Record: S21.COMP-0704 Issue 1 Dated 15 July 2010Master Documents List: S21.MDL-0704 Dated 5 July 2010.Flight Manual Supplement: S21.FMS-0062 Issue 1 Dated 15 July 2010.Supplement Manual: S21.TEC-0233 Issue 1 Dated 15 July 2010 or later rev (...) | None | 11.06.2012 |
| 10031459 | STC TWENTY ONE LIMITED | | | | F 406 | Installation of Thales I-MASTER A-Flight Radar Turret | EASA TCDS No EASA.A.109The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product (...) | STC Twenty One Limited Master Document List No. S21.MDL-0780 Issue 1 dated 13 August 2010STC Twenty One Limited Flight Manual Supplement No. S21.FMS-0063 or later revisions of the above listed documents approved by EASA | none | 17.08.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|--------------|----------------|----------------------|---|--|--|--|---|---------------|
| 10031662 | STC TWENTY ONE LIMITED | | | | CL-600-2B16 | Introduction of Swift Broadband | EASA.IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | STC Twenty One Ltd Master Document List S21.MDL-0786 Issue 1 dated 26 August 2010. STC Twenty One Ltd Flight Manual Supplement S21.FMS-0067 or later revisions of the above listed documents approved by EASA | 1. The Airworthiness Approval of the wireless system does not constitute Operational Approval for the use of transmitting PEDs on board of the aircraft. Operational Approval must be obtained from the responsible division of the operational authority of (...) | 02.09.2010 |
| 10031863 | STC TWENTY ONE LIMITED | | | | B757-200 | Introduction of Cabin Vu-123 Cabin Surveillance System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01/Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS" | Master Document List, doc. N° S21.MDL-0766 Issue 3 dated September 14th, 2010.- Compliance Record, doc N° S21.COMP-0766 Issue 2 dated June 9th (>,<)->2010. or later revisions of the above listed documents approved by EASA | None | 20.09.2010 |
| 10031961 | STC TWENTY ONE LIMITED | | | | BD-700-1A10 | Customized Aircraft Completion - ASN 9365 | EASA.IM.A.009The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRIF-01 "Installation of Wireless System" CRI D-01 "Installation of Interi (...) | STC Twenty One Ltd Master Document List No. S21.MDL-0795. STC Twenty One Ltd Airplane Flight Manual Supplement No. S21.FMS-0068. Instructions for Continued Airworthiness Midcoast Aviation Inc. Supplemental Maintenance Manual No. 35-027E225, or later rwi (...) | none | 27.09.2010 |
| 10032001 | STC TWENTY ONE LIMITED | | | | DC-9-11, DC-9-12, DC-9-13, DC-9-14, DC-9-15, DC-9-15F, DC-9-21, DC-9-31, DC-9-32, DC-9-32 (VC-9C), DC-9-32F, DC-9-32F (C-9A, C-9B), DC-9-33F, DC-9-34, DC-9-34F, DC-9-41, DC-9-51, DC-9-81 (MD-81), DC-9-82 (MD-82), DC-9-83 (MD-83), DC-9-87 (MD-87), MD-88, MD-90-30, 717-200 | Introduction of Cabin Vu-123 Cabin Surveillance System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-001/SC H-001 | Master Document List S21.MDL-0787 or later revisions of the above listed documents approved by EASA | None | 30.09.2010 |
| 10032098 | STC TWENTY ONE LIMITED | | | | BOMBARDIER CL600-2B19 | Removal of Divan Seating Restriction | EASA IM.A.023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | STC 21.MDL-0793 Issue 1 dated 30 September 2010. or later revisions of the above listed documents approved by EASA | This design change will allow the removal of an existing limitation on occupation of the forward most seating position on the installed divan (EASA STC No. EASA.A.S.03452 and EASA.A.S.03022 refers). If the arrester block installation assembly becomes unus (...) | 07.10.2010 |
| 10032232 | STC TWENTY ONE LIMITED | | | | CESSNA 206H, T206H | Introduction of Stand-By Battery Power Supply in accordance with S21.MDL-0770, issue 2, dated 14/10/2010 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Flight Manual Supplement S21.FMS-0070, issue 1 Installation Manual OAS-30-206-9, issue C, dated 06/10/2002. Instruction for Continued Airworthiness PAVCO document OAS-30-23-1529 or later revisions of the above listed documents approved by EASA | None | 19.10.2010 |
| 10033452 | STC TWENTY ONE LIMITED | | | | B757-200 | Carriage of Horses & Grooms in Freighter a/c | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. For changed areas these additional Special Conditions are part of the redefined Certification Basis, (...) | Master Document List, S21.MDL-0846, Initial Issue, 19/01/2011 Airplane Flight Manual Supplement, S21.FMS-0071, Issue 2, 19/01/2011 Instructions for Continued Airworthiness, S21.TEC-0286, Issue 1, 17/12/2010 or later revisions of the above listed documents approved by EASA | This design change is applicable to Boeing 757-200 aircraft converted under Boeing's Special Freighter Programme, approved by the UK CAA under Airworthiness Approval Note AAN 27308 (acceptable to EASA under the EASA Grandfather provisions). | 21.01.2011 |
| 10034641 | STC TWENTY ONE LIMITED | | REV. 3 | | PA-31 | Introduction of Chin Mount and re-located Pitot tubes. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List S21.MDL-0825 Issue 7 dated 23/05/2012; Flight Manual Supplement S21.FMS-0072 Issue 3; Compliance Record S21.COMP-0825 Issue 4 dated 23/05/2012; Instructions for Continued Airworthiness S21.TEC-0305 Issue 1; or later revisions of the above listed documents approved by EASA. | Not to be flown in icing conditions with the chin mount and aerodynamic fairing installed. The installation of a sensor turret is prohibited. Limitations as defined in AFMS S21.FMS-0072 Issue 3. | 29.05.2012 |
| 10034714 | STC TWENTY ONE LIMITED | | REV. 2 | | PA-31 | Introduction of Enlarged window and sensor turret provisions | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List S21.MDL-0914 Issue 5 dated 14/06/2013; Flight Manual Supplement S21.FMS-0077 Issue 2 dated 11/10/2013; Compliance Record S21.COMP-0914 Issue 2 dated 11/08/2013; Instructions for Continuing Airworthiness S21.TEC-0310 Issue 1 dated 2 (...) | Limited to A/C S/N 31-7712054 only. With a sensor turret installed, flight into known or forecast icing conditions is prohibited. Limitation as defined in AFMS. The installation of this modification by third persons is subject to written permission of th (...) | 14.06.2013 |
| 10034734 | STC TWENTY ONE LIMITED | | REV. 1 | | PA-31 | Introduction of a Belly Pod | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List S21.MDL-0915 Issue 3 dated 13/11/2012; Flight Manual Supplement S21.FMS-0080 Issue 2 dated 31/10/2012; Compliance Record S21.COMP-0915 Issue 2 dated 13/11/2012; Instructions for Continuing Airworthiness S21.TEC-0311 Issue 1 dated 20/04 (...) | Not to be flown in icing conditions with the belly pod installed. Limitations as defined in AFMS S21.FMS-0080 Issue 2 dated 31/10/2012 | 22.11.2012 |
| 10035173 | STC TWENTY ONE LIMITED | | | | B737-300 | Introduction of Number 2 Galley Assembly | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 25.1529 at Amdt 21 CRI H-1 - Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS | S21.MDL-0809 Issue 2 Master Document List dtd. 17.May 2011-S21.COMP-0809 Issue 2 Compliance Record dtd. 19.May 2011 S21.TEC-0316 Initial Issue 1 Supplemental Manual dtd. 18.May 2011 or later revisions of the above listed documents approved by EASA | None | 01.06.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|------------------|----------------|----------------------|---|---|--|--|---|---------------|
| 10035193 | STC TWENTY ONE LIMITED | | | | EMB-145 | Introduction on EMB-145 of Saftglo photoluminescent FPEEPMS | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List doc. S21.MDL-0842 dated 24.05.2011 Aircraft Flight Manual Supplement doc. S21.FMS-0075 Issue 1 Compliance Record doc. S21.COMP-0842 dated 24.05.2011 or later revisions of the above listed documents approved by EASA | none | 03.06.2011 |
| 10035344 | STC TWENTY ONE LIMITED | | | | A320-214 | Custom Interior A320 Jubilee | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The following EASA Certification Review Items (CRIs) are part of the certification basis (...) | STC Twenty One Limited documents:- Master Document List S21.MDL-0909 initial issue dated 02nd June 2011.- Aeroplane Flight Manual (AFM) Supplement S21.FMS-0079 Issue 1.- For documents and supplement manuals related with instructions for continued air (...) | This design change approval is applicable on Aircraft model Airbus A320-214, aircraft serial number 4199. | 16.06.2011 |
| 10035624 | STC TWENTY ONE LIMITED | | | | BD-700-1A11 | Customized Interior Completion - ASN 9308 | EASA.IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | STC Twenty One Ltd Master Document List No S21.MDL-0794 Issue 1 STC Twenty One Ltd Flight Manual Supplement No. S21.FMS-0084 Issue 1 Midcoast Aviation Inc. Instructions for Continued Airworthiness No 35-027E205 or later revisions of the above listed documents approved by EASA | Restricted to ASN 9308 | 05.07.2011 |
| 10036234 | STC TWENTY ONE LIMITED | | | | 767-300 | Omnibus Modification for the Approval of Air New Zealand Engineering Orders Embodied on the A/C | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this approval. | Master Document List, S21.MDL-0810, Issue 1, 15/07/2011 or later revisions of the above listed documents approved by EASA | none | 24.08.2011 |
| 10036644 | STC TWENTY ONE LIMITED | | | | CL-600-2B16 | Introduction of Corporate Interior | EASA.IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | STC Twenty One Master Document List No. S21.MDL-0954 Initial Issue, dated 22 September 2011 Midcoast Aviation AFMS No. 34-424M005 Midcoast Aviation AFMS No. 34-261M017 or later revisions of the above listed documents approved by EASA | Restricted to ASN 5616 | 26.09.2011 |
| 10036793 | STC TWENTY ONE LIMITED | | | | B777-200, B777-200LR, B777-300, B777-300ER, B777F | Introduction of Cabin Vu 123 Cabin Surveillance System. | EASA.IM.A.003 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Generic CRI H-01 EWIS The requirements for environmental protection and th (...) | STC Twenty One Ltd Master Document List No. S21.MDL-0857 Issue 2, dated 09 September 2011 or later revisions of the above listed documents approved by EASA. | None. | 06.10.2011 |
| 10037523 | STC TWENTY ONE LIMITED | P-EASA.A.S.03880 | | | A318, A319, A320, A321 | Introduction of an Avionics Mini Quick Access Recorder MK III with Cellular Module | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. 1. Special Condition SC H-01 'Enhanced Airworthiness Programme for Aeroplane Systems - I (...) | Certification Programme, ref. S21.PROG-0639, Issue 3, dated 13th October 2011- Master Document List, ref. S21.MDL-0638, Issue 2, dated 13th October 2011- Compliance Checklist, ref. S21.CCL-0638, Issue 2, dated 14th October 2011- Equipment Approval, r (...) | Not applicable | 02.12.2011 |
| 10039188 | STC TWENTY ONE LIMITED | | | | A319-115 | Introduction of Custom Interior | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The following EASA Certification Review Items (CRIs) are part of the certification basis (...) | STC Twenty One Limited documents:- Master Document List S21.MDL-0868 issue 1, dated 13th April 2012.- Aeroplane Flight Manual (AFM) Supplement S21.FMS-0090 Issue 1, dated 13th April 2012.- For documents and supplement manuals related with instructions f (...) | This design change approval is applicable on Aircraft model Airbus A319-115, aircraft serial number 4622. Any subsequent change to the modes of operation of the WLAN approved under this design change will require a re-examination to determine the compli (...) | 16.04.2012 |
| 10039812 | STC TWENTY ONE LIMITED | | REV. 2 | | B737 | Activation of a Controller-Pilot Data-Link Communications (CPDLC) capability on Spectralux "D-LINK" ACARS VDL Mode 2 transceiver. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this approval.- CRI F-1 'Datalink services for the Single European Sky' issue 2 dated 24 May 2012.- CRI H-1 'Enhance (...) | Certification Programme S21.PROG-0813, Issue 3, dated 12 February 2012- Compliance Check List, S21.CCL, Issue 2, dated 22 February 2012- Master Data List S21.MDL-0813, Issue 1, dated 19 April 2012- AFM Supplement S21.FMS-0069, Issue 3, dated 7 March 2012 (...) | This CPDLC installation is only intended to be used for routine datalink exchanges during enroute operations in upper airspace. Prior to installation of this STC it must be determined that the interrelationship between this modification and any other (...) | 16.04.2013 |
| 10040493 | STC TWENTY ONE LIMITED | | | | CL-600-2A12 | Introduction of Swift Broadband | EASA.IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged (...) | STC Twenty One Ltd Master Document List No. S21.MDL-0879 Issue 3 Flight Manual Supplement No. S21.FMS-0078 Issue 1 or later revisions of the above listed documents approved by EASA | none | 06.07.2012 |
| 10040691 | STC TWENTY ONE LIMITED | | REV. 1 | | PA-31 | Provisions for the installation of a Sensor Turret | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List S21.MDL-1027 Issue 3 dated 18/04/2013 Flight Manual Supplement S21.FMS-0091 Issue 2 dated 18/04/2013 Compliance Record S21.COMP-1027 Issue 3 dated 05/06/2013 Instructions for Continuing Airworthiness S21.TEC-0374 Issue 1 dated 18/07 (...) | With a sensor turret installed, flight into known or forecast icing conditions is prohibited. Limitations as defined in AFMS S21.FMS-0091 Issue 2 The installation and subsequent operation of a sensor turret requires separate approval. Prior to installa (...) | 07.06.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|----------------|----------------|----------------------|-----------------------------|--|--|---|--|---------------|
| 10041363 | STC TWENTY ONE LIMITED | | | | B757-200/-200PF/-300 | Introduction of Antenna Mounting Provision | The Certification Basis for the original product remains applicable to this certificate/ approval. | S21.MDL-1029 Issue 1S21.ACC-0470 Issue 1S21.TEC-0349 Issue 1S21.COMP-1029 Issue 153-15-10108 Issue A23-00-10256 Issue A23-15-10346 Issue A23-15-10335 Issue A or later revisions of the above listed documents approved by EASA | None | 11.09.2012 |
| 10041598 | STC TWENTY ONE LIMITED | | | | BOEING 767-200/-300 | Introduction of Iridium Antenna. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | STC Twenty One Ltd Master Document List No. S21.MDL-1080 Issue 3, or later revisions of the above listed documents approved by EASA. | None. | 01.10.2012 |
| 10043012 | STC TWENTY ONE LIMITED | EASA.A.S.01124 | | | CAP 232 | Introduction of Rudder, Throttle and Brake Controls for Hand Operation by Paraplegic Pilot with upper Body Strength only | | Master Document List S21.MDL-0557 Issue 3 dated 02/10/2008 | 1., Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2., Serial Number 11 only. | 16.03.2009 |
| 10043085 | STC TWENTY ONE LIMITED | EASA.A.S.03748 | | | 737-300 | Carriage of Horses & Grooms | | Master Document List S21.MDL-0612 Issue 1, Airplane Flight Manual Supplement S21.FMS-0051 Issue 1, Instructions for Continued Airworthiness S21.TEC-0203 Issue 1 or later approved revisions of those documents. | 1., This design change is applicable to Boeing 737-300 aircraft converted under FAA STC SA2968SO, SA2969SO, and SA2970SO& SA2971SO as previously been validated by the UK CAA under Airworthiness Approval Note AAN 24984. | 18.05.2009 |
| 10043462 | STC TWENTY ONE LIMITED | | | | LEARJET MODEL 60 | STC Twenty One Limited Project No. S21.53115-1037 Installation of Satcom Antenna This design change introduces a satcom antenna to the crown skin of the aircraft. The design change is limited to the mechanical changes to the aircraft only. The antenna will (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | STC Twenty One Limited, Master Drawing List S21.MDL-1073 Issue 1, dated 30 January 2013 STC Twenty One Limited (->-) Supplement Manual S21.TEC-0371, Issue 1, dated 28 January 2013 or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 30.01.2013 |
| 10043472 | STC TWENTY ONE LIMITED | | | | DHC-8-401, DHC-8-402 | Introduction of CabinVu 123 Cabin Surveillance System. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation must be in accordance with STC Twenty One Master Document List S21.MDL-0626, Issue 3, Maintenance must be in accordance with Supplement Manual S21.TEC-0278, Issue 3, Operation must be in accordance with AD Aerospace Operating Manual CabinVu (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 31.01.2013 |
| 10044177 | STC TWENTY ONE LIMITED | | REV. 1 | | MYSTERE FALCON 900 | Upgrade To Existing VIP Interior. This Rev. 1 is to amend a typographical error in "Description of Design change". | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - Special Condition(s): D-05 Glass Screens of Displays and Monitors The requirements for environmental protection and the (...) | STC Twenty One Master Document List No. S21.MDL-1118 Initial Issue, dated 21 March 2013, Instructions for Continued Airworthiness, Jet Aviation Saint-Louis Inc Instructions for Continued Airworthiness document references 14-027E061, or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 25.03.2013 |
| 10044445 | STC TWENTY ONE LIMITED | | | | A330-200, -200F, -300 | Introduction of VHF Comm & Satcom Antennas. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | STC Twenty One Limited Master Document List No. S21.MDL-0998 Initial Issue dated 07 March 2013, Instructions for Continued Airworthiness Supplement Manual No. S21.TEC-0370, or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 15.04.2013 |
| 10044818 | STC TWENTY ONE LIMITED | | | | B737-300 | Introduction of replacement ELT Antenna onto repaired fuselage crown skin FS 833, Str 1 to 2R. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Compliance checklist S21.CCL-1129 dated May 02nd, 2013 Master Document List S21.MDL-1129 dated May 02nd, 2013 Compliance record S21.COMP-1128 dated April 30th, 2013 Technical Report S21.TREP-1722 dated April 30th, 2013 or later revisions of the above listed documents approved by EASA. | Applicability limited to aircraft Serial Number 29072 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 07.05.2013 |
| 10044909 | STC TWENTY ONE LIMITED | | | | 737-800 | Introduction of SatLINK antenna in accordance with Archeion Holdings LLC. Iridium Antenna Installation Subkit 2604773K-1-101 Revision H | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Report 26047-ICA Rev F: Instructions for Continued Airworthiness for the installation of an Avionics Inc. SatLINK communication management system with cabin wireless capability in a Boeing 737-300/-400/-700/-800 series aircraft or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 17.05.2013 |
| 10045862 | STC TWENTY ONE LIMITED | | | | 757-200, 757-200PF, 757-300 | Introduction of TCAS 2000 v7.1 | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Document List - S21.MDL-1105 Issue 1 Flight Manual Supplement - S21.FMS-0096 Issue 1 Flight Manual Supplement - S21.FMS-0098 Issue 1 or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 24.07.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------|----------------|----------------------|--|--|--|---|--|---------------|
| 10046675 | STC TWENTY ONE LIMITED | | | | 737-700 | Introduction of Corporate VIP Interior | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): D-02 Isolated Compartments-D-05 Glass Screens of Displays and Monitors-D-09 Injury Criteria for Side-Fa (...) | STC Twenty One Ltd Master Document List No. S21.MDL-1081 Issue 2STC Twenty One Ltd Flight Manual Supplement No. S21.FMS-0100 Issue for later revisions of the above listed documents approved by EASA | None | 10.10.2013 |
| 10015317 | S-TEC CORPORATION | EASA.IM.A.S.00244 | | FAA STC SA5321SW-D | BOEING 767-300ER | Reconfiguration of passenger cabin interior from 204 passengers to 250 passengers in accordance with DER Services Inc. Master Document List (MDL) Report No. MDL-A2004-026-1 dated 15 June 2004, or later EASA approved revisions | | | 1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated.2. If transfer of this Certificate is requested under Part 21A.116, the STC (...) | 14.10.2004 |
| 10015320 | S-TEC CORPORATION | EASA.IM.A.S.00268 | REV. 1 | FAA STC SA09290AC-D | PIPER PA-32-260 PIPER PA-32-300 | Installation of S-TEC System 20/30 Single Axis and Two Axis Automatic Flight Guidance Systems Model ST-733-20/30 SA09290AC-D. | | Bulletin No.833 dated October 24, 1997 and Master Drawing List No.921034 dated October 24, 1997 or later FAA Approved Revision Flight Manual Supplement: Airplane Flight Manual Supplement P/N 891554 dated October 29, 1997 or later FAA Approved Revision (...) | 1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated.2. If transfer of this Certificate is requested under Part 21A.116 the STC will (...) | 10.09.2008 |
| 10015321 | S-TEC CORPORATION | EASA.IM.A.S.00269 | REV. 2 | FAA STC SA8906SW-D | PA-28-151, PA-28-161 PA-28-181, PA-28-201T PA-28-236 | Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System Model ST-568. | The Certification Basis for the original product remains applicable to this certificate/ approval | FAA/DAS Approved S-TEC Bulletin No. 668 Revision 3, dated 21 July 2001 or later revision- FAA/DAS Approved S-TEC Master Drawing List No. 92748 Revision C, dated 21 July 2001 or later revision- FAA/DAS Approved Pilot's Operating Handbook and/or Airpla (...) | None | 14.09.2010 |
| 10015326 | S-TEC CORPORATION | EASA.IM.A.S.00281 | REV. 1 | FAA STC SA09386AC-D | GRUMMAN AA-5 GRUMMAN AA-5A GRUMMAN AA-5B | Installation of S-TEC System 30 Single and Two Axis Automatic Flight Guidance System, Model ST-840-30, FAA STC SA09386AC-D | | Bulletin No 940 dated September 14 1998 and Master Drawing List No 921115 dated September 14 1998 or later FAA Approved Revision Flight Manual Supplement: FAA/DAS Approved Supplemental Flight Manual, Part Number 891787 dated September 17 1998 or late (...) | 1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated.2. If transfer of this certificate is requested under Part 21A.116, the STC will be reissued | 11.09.2008 |
| 10015327 | S-TEC CORPORATION | EASA.IM.A.S.00289 | | FAA STC SA09264AC-D | CESSNA F182-Q | Installation of S-TEC System 30 Two Axis Automatic Flight Guidance System, Model ST-718/30, according to Bulletin No. 818, dated July 30, 1997 and Master Drawing List No. 921016, dated July 30, 1997 or later FAA approved revisions of the above data (...) | | | 1. EASA Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement, dated October 21, 2004 is required for Reims-Cessna Model F182Q for S-TEC System 30 or later EASA approved revisions of the above supplement.2. Compatible (...) | 28.10.2004 |
| 10015362 | S-TEC CORPORATION | EASA.IM.A.S.00676 | REV. 2 | FAA STC SA09254AC-D | PIPER PA-28-140 PIPER PA-28-150 PIPER PA-28-160 PIPER PA-28-180 PIPER PA-28-235 | Installation of S-Tec System 20/30 Single and Two Axis Automatic Flight Guidance Systems, Model ST-711-20/30 According FAA STC SA09254AC-D | | Bulletin No. 811, dated 01.07.1997 and Master Document List No 921011, dated 01.07.1997 or later approved revision (14 Volt System) FAA approved POH and/ or AFM- Supplement P/N 891448, dated 07.07.1997 or later approved Revision for S-TEC System 20 on (...) | 1. Limitation according FAA STC SA09254AC-D.2. Compatibility of this modification with other previously approved modifications must be determined by the installer.3. This STC is approved only for the product configuration as defined in the approved des (...) | 11.09.2008 |
| 10015381 | S-TEC CORPORATION | EASA.IM.A.S.01066 | REV. 1 | FAA STC SA8986SW-D | CESSNA 210L CESSNA 210M CESSNA 210N CESSNA 210R CESSNA T210L CESSNA T210M CESSNA T210N CESSNA T210R | Installation of S-TEC System 55/55X Two Axis Automatic flight Guidance System, Model ST-566, according to Bulletin No 666, Rev 2 dated 1-25-01 or later approved revisions | | -S-TEC Master Drawing List No 92734, Rev B, dated 1-25-01, or later approved revision- FAA/DAS approved Supplements to pilot's operating Handbooks and/or Approved Airplane Flight Manual, as defined in the Limitations and Conditions section of FAA STC (...) | 1. As defined in the Limitations and Conditions section of FAA STC SA8986SW-D, latest approved revision.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other prev (...) | 09.09.2008 |
| 10015383 | S-TEC CORPORATION | EASA.IM.A.S.0107 | REV. 1 | | CESSNA 177 CESSNA 177A CESSNA 177B | Installation of S-TEC System 20/30 Single and Two-axis Automatic Flight Guidance Systems. | | Installation of S-TEC System 20/30 Single and Two-axis Automatic Flight Guidance Systems. Model ST-741-20/30. In accordance with Bulletin No 841, dated 11-03-97 and Master Drawing list No 921040, dated 11-03-97 or later FAA Approved revisions as de (...) | 1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated.2. If transfer of this Certificate is requested under Part 21A.116, the STC (...) | 10.09.2008 |
| 10015397 | S-TEC CORPORATION | EASA.IM.A.S.01099 | | FAA STC SA09502AC-D | AVIATION AIRCRAFT A-1 AVIATION AIRCRAFT A-1A AVIATION AIRCRAFT A-1B | Installation of S-TEC System 30 ALT Altitude Hold System, Model ST-881 | | FAA STC SA09502AC-D-Bulletin No 981 Revision 1, dated September 14, 2000 and Master Drawing List no 921193, Revision A, dated September 14, 200 or later FAA approved revision (14/28 Volt Systems) FAA/DAS Approved Airplane Flight Manual Supple (...) | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 01.03.2005 |
| 10015419 | S-TEC CORPORATION | EASA.IM.A.S.01142 | REV. 1 | FAA STC SA09130AC-D | PILATUS PC 6 SERIES | Installation of Meggitt Avionics S-TEC Autopilot 55/55X | | Master Drawing List ST-598 P/N 92824 Rev C or later approved revisions DWG No 55603 applies for control stick modification Airplane Flight Manual Supplement S-TEC P/N 891229S rev 2 or later approved revisions Instructions for Continued Airworthiness P/N 81581 Rev 2 or later approved revisions | 1. Limited to PC-6/B1-H2; PC-6/B2-H2; PC-6/B2-H4 and PC-6/B2-H4 when modified with underwing fuel tanks.2. If stabilizer electrical trim interrupt switch S043 is installed, the optional trim wiring in accordance with detail "A" on S-TEC drawing No 992 (...) | 10.09.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------|-------------------|----------------|----------------------|--|--|---------------------|--|--|---------------|
| 10015441 | S-TEC CORPORATION | EASA.IM.A.S.01181 | | FAA STC SA09018AC-D | CESSNA A185F | Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-596. Bulletin No.696 Revision 3 dated February 2 2001 & Master Drawing List No. 92801 Revision C dated February 2 2001 or later approved revision (28 volt system) (...) | | Bulletin No 696 Revision 3 dated February 2 2001 & Master Drawing List No 92801 revision C dated February 2 2001 or later approved revision (28 volt system)S-TEC AFM Supplement P/N 891191 & 891192 dated August 25 1995 or later approved revision | Cessna Model A185F & Cessna Model A185F when equipped with STC SA1441WE (Robertson STOL) or STC SA968CE (Horton STOL) and/or STC SA2529WE (Internal Tip Tanks) | 25.07.2005 |
| 10015442 | S-TEC CORPORATION | EASA.IM.A.S.01182 | | FAA STC SA09195AC-D | COMMANDER 112, 112B, 112TC, COMMANDER 112TCA, 114, 114A | Installation of S-TEC System 55/55X wo Axis Automatic Flight Guidance System. Model ST-658, in accordance with FAA STC SA09195AC-D. | | FAA/DAS approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement, P/N 891326, Revision 2, dated 5-23-01 is required for:1. Commander Aircraft models 112, 112B,112TC, 112TCA, 114 and 114A.2. Commander Aircraft models 112 and 112B when modified per STC SE5203NM (turbocharger/interc | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 28.04.2005 |
| 10015455 | S-TEC CORPORATION | EASA.IM.A.S.01201 | | FAA STC SA09283AC-D | NAVION A (L-17B) NAVION B NAVION G NAVION H | Installation of S-TEC System 20/30 Single and Two Axis Automatic Flight Guidance System Model ST-724-20/30 in accordance with FAA STC SA09283AC-D | | -FAA/DAS approved Supplemental Flight Manual P/N 891535, Rev 1 for Navion models A, B, D, E, F, G and H when modified by a/c Alto Airport Inc or Symon's Engineering Horizontal Stabilizer Angle of Incidence Change with or without tip tanks and the ab (...) | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 30.05.2005 |
| 10015478 | S-TEC CORPORATION | EASA.IM.A.S.01232 | REV. 1 | FAA STC SA5112SW-D | PIPER PA-30 PIPER PA-39 | Installation of S-TEC System 60 Two Axis Automatic Flight Guidance System Model ST-127. FAA STC SA5112SW-D | | Master Drawing List No. 92139 Rev. G dated 06-01-97. Airplane Flight Manual Supplement PN 89157 Rev. 3 dated 23-01-97 or later approved revisions. | 1. This STC is also eligible on the concerned model when modified per STC SA727WE.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other a (...) | 09.09.2008 |
| 10015555 | S-TEC CORPORATION | EASA.IM.A.S.01452 | REV. 1 | FAA STC SA09147AC-D | CESSNA 152 CESSNA A152 | Installation of S-TEC System 40/50 Single and Two Axis Automatic Flight Guidance System. Model ST-628, according to Bulletin No 728, dated 3-29-96 or later approved revisions | | -S-TEC Master Drawing List No 92921, dated 3-29-96, or later approved revisions-FAA/DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement, P/N 891246, dated 4-03-96 or later approved revisions is required for Cessna Models (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 10.09.2008 |
| 10015556 | S-TEC CORPORATION | EASA.IM.A.S.01453 | REV. 1 | FAA STC SA09148AC-D | CESSNA F152 CESSNA FA-152 | Installation of S-TEC System 40/50 Single and Two Axis Automatic Flight Guidance System. Model ST-628, according to Bulletin No 728, dated 3-29-96 or later approved revisions | | -S-TEC Master Drawing List NO 92921, dated 3-29-96, or after approved revisions-FAA/DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement, P/N 891247, dated 4-03-96 or later approved revisions is required for Reims Aviation (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 10.09.2008 |
| 10015557 | S-TEC CORPORATION | EASA.IM.A.S.01454 | REV. 1 | FAA STC SA5302SW-D | CESSNA 210 | Installation of S-TEC System 40/50 Single and Two Axis Automatic Flight Guidance System. Model ST-212-40/50, according to Bulletin No 312, dated 3-21-88 or later approved revisions | | -S-TEC Master Drawing List No 92234, rev D, dated 3-21-88, or later approved revisions-FAA Approved Supplements to Pilot's Operating Handbooks and/or FAA Approved Airplane Flight Manual, as defined in the Limitations and Conditions section of FAA STC (...) | 1. As defined in the Limitations and Conditions section of FAA STC SA5302SW-D, latest approved revision2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previ (...) | 09.09.2008 |
| 10015558 | S-TEC CORPORATION | EASA.IM.A.S.01455 | REV. 1 | FAA STC SA09628AC | TWIN COMMANDER AIRCRAFT 690C TWIN COMMANDER AIRCRAFT 690D TWIN COMMANDER AIRCRAFT 695 TWIN COMMANDER AIRCRAFT 695A | Installation of S-TEC/Meggitt MAGIC 01298-07-01-005 Digital Flight Control SystemFAA STC SA09628AC | | S-TEC Master Drawing List No. 921128, Rev. 5, dated 12 December 2003- Airplane Flight Manual Supplement No. 892134, dated 17 December 2003- S-TEC Instruction for Continued Airworthiness No. 81683 Rev. 3 | Airplane with S-TEC/Meggitt 01298-07-01-005 Digital Flight Control System is prohibited from flight into known icing condition. STC SA01627CH, Meggit EFIS installation is required for this modification. Compatibility of this design with previously ap (...) | 17.10.2006 |
| 10015559 | S-TEC CORPORATION | EASA.IM.A.S.01456 | | FAA STC SA5242SW-D | AERMACCHI F 260 B AERMACCHI F 260 C AERMACCHI F 260 D | Installation of S-TEC System 40/50, Single and Two Axis Automatic Flight Guidance Systems Model ST-235-40/50FAA STC SA5242SW-D | | *Master Drawing List No. 92261 Rev. C dated 21.3.91 | 1. Only eligible on model F260B when equipped with Aileron Assemblies P/N SF260-02-05-01 (L.H.) and P/N SF260-02-05-02 (R.H.)2. This STC is approved only for the product configuration as defined in the approved design data referred to in the p (...) | 13.03.2006 |
| 10015588 | S-TEC CORPORATION | EASA.IM.A.S.01499 | | FAA STC SA09239AC-D | CESSNA F172-N | Installation of S-TEC System 20/30 Single and low Axis Automatic Flight Guidance Systems | | Bulletin No 774, dated 11 June 1997 and Master drawing listNo 921007, dated 11 June 1997 or later FAA Approved revision of the above data.EASA approved Pilot's Operating Handbookand/or Airplane Flight Manual Supplement P/N EASA891427, dated 14 June (...) | Also eligible on the above models when modified by 180 HP Engine conversion. EASA approval of this STC must be checked.This STC is issued as result of the validation of the S-TEC Corporation STC No SA09239AC-D approved by the FAA on June 26, 1997. | 04.08.2005 |
| 10015604 | S-TEC CORPORATION | EASA.IM.A.S.01540 | REV. 1 | FAA STC SA7209SW-D | CESSNA 337D CESSNA 337E CESSNA 337F CESSNA 337G CESSNA 337H CESSNA P337H CESSNA T337D CESSNA T337E CESSNA T337F CESSNA T337G CESSNA T337H CESSNA T337H-SP | Installation of S-TEC System 60 Two Axis Automatic Flight Guidance System with Optional Autotrim System, Model ST-314, according toBulletin No 414, dated 3-14-90 or later approvedrevision | | -S-TEC Master Drawing List No 92503, dated 3-14-90 or laterapproved revision-FAA/DAS Approved Supplemental Flight Manual, Part No 89845, dated 4-06-90 is required for Cessna Models 337D, T337D, 337E, T337E, 337F, T337F, 337G, T337G, 337H,P337H, T337 (...) | 1. Eligible when above models are modified per FAA STC SA1527WE (Robertson STOL Conversion)2. Eligible when the above models are modified per FAA STC SA937CE (Horton STOL Craft Conversion)3. It is the operator's responsibility to ensure theabove ment (...) | 12.09.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------|-------------------|----------------|----------------------|--|--|---------------------|--|--|---------------|
| 10015606 | S-TEC CORPORATION | EASA.IM.A.S.01541 | REV. 1 | FAA STC SA7210SW-D | REIMS AVIATION F-337E REIMS AVIATION F-337F REIMS AVIATION F-337G REIMS AVIATION F-337H REIMS AVIATION FT-337E REIMS AVIATION FT-337F REIMS AVIATION FT-337GP REIMS AVIATION FT-337HP | Installation of S-TEC System 60 Two Axis Automatic Flight Guidance System with Optional Autotrim System, Model ST-314, according to Bulletin No 414, dated 3-14-90 or later approved revisions. | | 1. S-TEC Master Drawing List No 92503, dated 3-14-90, or later approved revisions. 2. FAA/DAS Approved Supplement Flight Manual, Part No 89847, dated 4-06-90 is required for Reims Aviation S.A. Models F337E, FT337E, F337F, FT337F, F337G, FT337GP (...) | 1. Eligible when the above models are modified per FAA STC SA1627WE (Robertson STOL Conversion). 2. Eligible when the above models are modified per FAA STC SA937CE (Horton STOL Craft Conversion). 3. It is the operators responsibility to ensur (...) | 09.09.2008 |
| 10015633 | S-TEC CORPORATION | EASA.IM.A.S.01589 | | FAA STC SA7805SW-D | PIPER PA-28R-180 PIPER PA-28R-200 PIPER PA-28R-201 PIPER PA-28R-201T PIPER PA-23 PIPER PA-23-160 | Installation of S-TEC Manual Trim System (field installation) i.a.w. MDL 92549 Rev A | | -FAA STC SA7805SW-D-AFM Supplement P/N 89940-IFCA, P/N 81322, Rev 2 | Limited to installations without the trim monitor system | 30.01.2006 |
| 10015634 | S-TEC CORPORATION | EASA.IM.A.S.01590 | | FAA STC SA7815SW-D | PIPER PA-23 PIPER PA-23-160 | STC SA7815SW-D, Installation of S-TEC System 60 Two Axis Automatic Flight Guidance System, Model ST-486 | | FAA STC SA7815SW-DFAA AFMS P/N 89968, Rev. 2,IFCA P/N 81431 | 1. Eligible on models modified per STC SA504WE and/or STC SA824WE 2. Eligible on models modified per STC SA461SW, SA636SW, SA671SW and 899SW3. Instructions for continued airworthiness acc. Doc. P/n 81431 5. Maintenance Instructions 2-4. must be per (...) | 17.02.2006 |
| 10015669 | S-TEC CORPORATION | EASA.IM.A.S.01645 | REV. 1 | FAA STC SA09214AC-D | CESSNA 172R CESSNA 172S | Installation of S-TEC System 30 ALT Altitude Hold System, Model ST-673, according to Bulletin No 773, Revision 4 dated 9-22-98 or later approved revision | | - S-TEC Master Drawing List No 92991, Revision D dated 9-22-98 or later approved revisions- FAA/DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement, P/N 891376, Revision 1, dated 9-25-98 or later approved revisions-Inst (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 10.09.2008 |
| 10015783 | S-TEC CORPORATION | EASA.IM.A.S.01839 | | FAA STC SA09522AC-D | BEECH 95 BEECH B95A BEECH D95A BEECH E95 | Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-898. | | Installation according Bulletin No.998, dated 2-12-02 and Master Drawing List No. 921214, dated 2-12-02 or later FAA Approved revisions of the above data (28V System). | FAA/DAS Approved Pilot's Operating Handbook and / or FAA Approved Airplane Flight Manual Supplement P/N 892113 dated 2-18-02 or later approved revision, is required for Beech models 95 and B95 FAA/DAS Approved Pilot's Operating Handbook and / or FAA (...) | 17.10.2006 |
| 10015784 | S-TEC CORPORATION | EASA.IM.A.S.01840 | | FAA STC SA8908SW-D | CESSNA 335 CESSNA 340 CESSNA 340A | Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-575 | | Installed in accordance with Bulletin No 675, Revision 3, dated 5-24-01 and Master Drawing List No 92752, Revision C, dated 5-24-01 or later approved revision.Operation in accordance with FAA/DAS Pilot's Operating Handbook and/or Airplane Flight Manu (...) | Eligible to be used on aircraft 340 and 340A when modified per FAA STC SE4327SW (RAM Engine Conversion only if this STC has been EASA approved. | 26.09.2006 |
| 10015786 | S-TEC CORPORATION | EASA.IM.A.S.01842 | REV. 1 | FAA STC SA09000AC-D | VULCANAIR P68 B VULCANAIR P68 C VULCANAIR P68 C-TC | Installation of S-TEC System 55 two axis automatic flight guidance system, model ST-585, in accordance with FAA STC SA 09000AC-D | | FAA/DAS approved POH/AFM Supplement for Partenavia P68B, P68C and P68C-TC with S-TEC System 55 two axis automatic flight guidancesystem P/N 891160, Revision 1 or later FAA/DAS approved revisions | 1. Also eligible on the above models with Camera Port2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 10.09.2008 |
| 10015801 | S-TEC CORPORATION | EASA.IM.A.S.01866 | | FAA STC SA09527AC-D | TWIN COMMANDER AIRCRAFT 690 TWIN COMMANDER AIRCRAFT 690A TWIN COMMANDER AIRCRAFT 690B | Installation of S-TEC Digital Flight Control System (autopilot) Part Numbers 01284 or 01298-01-01-002 or 01-304-04-01-008, (MAGIC 2100) (ST-902) in Twin Commander 690, 690A and 690B, also if modified by any of the following STCs: STC SA236CH (Garre (...) | | Master Data List No. 921219, Rev. B or later FAA approved revision.Installation Instructions ST-902-1, Rev. C, or later released revisionFAA/DAS approved Airplane Flight Manual Supplement No. 89123, Rev. 3, or later approved revisionInstructions for (...) | If airplane is modified by STC SA236CH (Garrett TPE 331-10T-515 engines), STC SA3861NM ("Q" tip propellers), or STC SA1359GL (winglets), installer must check availability of EASA-approval. | 20.10.2006 |
| 10015802 | S-TEC CORPORATION | EASA.IM.A.S.01867 | REV. 1 | FAA STC SA09007AC-D | CESSNA 182P CESSNA 182Q | Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System Model ST-587. | | In accordance with Bulletin No 687, Revision 2, dated 2-15-01 and Master Drawing List No 92785, Revision B, dated 2-15-01 or later approved revisions.FAA/DAS Approved Supplement Flight Manual, Part No 891171, dated 6-01-95 is required for Cessna Mod (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engineconfigurations shall be determined by the installer. | 10.09.2008 |
| 10015804 | S-TEC CORPORATION | EASA.IM.A.S.01869 | | FAA STC SA09008AC-D | CESSNA F182-P CESSNA F182-Q | Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System Model ST-587. | | In accordance with Bulletin No. 687 Revision 2, dated 2-15-01 and Master Drawing List No. 92785, Revision B, dated 2-15-01 or later approved revisions.FAA/DAS approved Supplement Flight Manual, Part No. 891173, dated 6-01-95 is required forReims Cess (...) | | 26.04.2006 |
| 10015827 | S-TEC CORPORATION | EASA.IM.A.S.01916 | | FAA STC SA5125SW-D | PIPER PA-28R-180 PIPER PA-28R-200 PIPER PA-28R-201 PIPER PA-28R-201T | Installation of S-TEC System 60 Two Axis Automatic Flight Guidance System, Model ST-137, with Optional Autotrim System. | | Bulletin No. 237, Rev. 2 or later approved revision,Master Drawing List No. 92160, Rev. B, dated 6-04-90, or later approved revisionAFM-Supplements as mentioned below Instructionsfor Continued Airworthiness, P/N 81217, Rev. 1 or later | 1. FAA Approved Supplement to Pilot's Operating Handbook and / or FAA Approved Airplane Flight Manual, P/N 89176-1, dated 06-21-90, or later approved revision, is required for Piper Models PA-28R-201 and PA-28R-201T. 2. FAA Approved Supplement to Pilot (...) | 11.10.2006 |
| 10015901 | S-TEC CORPORATION | EASA.IM.A.S.02027 | | FAA STC SA9232AC-D | CESSNA 210G CESSNA 210H CESSNA 210J CESSNA T210G CESSNA T210J | Installation of S-TEC System 20/30 Single and Two Axis Automatic Flight Guidance System, Model ST-687-20/30 | | Bulletin No. 787, dated 27 May 1997 and Master Drawing ListNo. 921000, dated 27 May 1997 or later approved revisions of the above data (14 Volt System) | 1. Also eligible on Models T210G, T210H and T210J when modified per FAA STC SA2689SW (RAM Modification)2. FAA/DAS Approved Supplemental Flight Manual, P/N 891412, dated 30 May 1997, is required for Cessna Models 210G, T210G, 210H, T210H, 210J and T2 (...) | 27.06.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------|-------------------|----------------|----------------------|---|--|---------------------|---|--|---------------|
| 10015902 | S-TEC CORPORATION | EASA.IM.A.S.02028 | | FAA STC SA09237AC-D | CESSNA 172M CESSNA 172N | Installation of S-TEC System 20/30 Single and Two Axis Automatic Flight Guidance Systems, Model ST-674-20/30 | | Bulletin No. 774, dated 11 June 1997 and Master Drawing List No. 921007, dated 11 June 1997 or later approved revisions of the above data (14 Volt System). | 1) Also eligible on the above models when modified by 180 H.P. Engine Conversion.2) FAA/DAS Approved Supplemental Flight Manual, P/N 891423, dated 16 June 1997 is required for S-TEC System 20 for Cessna Model 172M and the above model when modified b (...) | 26.06.2006 |
| 10015905 | S-TEC CORPORATION | EASA.IM.A.S.02031 | | FAA STC SA09313AC-D | BEECH A35 BEECH B35 BEECH C35 BEECH D35 BEECH E35 BEECH F35 BEECH G35 | Installation of S-TEC System 20/30 Single and Two Axis Automatic Flight Guidance Systems, Model ST-757-20/30 | | Bulletin No. 857, dated 12-01-97 and Master Drawing List No. 921052, dated 12-01-97 or later approved revisions of the above data (14 Volt System) | Also eligible on Model 35 when modified per STC SA1-460 (Auxiliary Wing Tip Fuel Tanks).FAA/DAS approved supplemental flight manual, P/N 89160, dated 12-05-97 is required for Beech model 35 and this model when modified per STC SA1-460 (Auxiliary Win (...) | 12.06.2006 |
| 10015922 | S-TEC CORPORATION | EASA.IM.A.S.02055 | | FAA STC SA09009AC-D | CESSNA 182E CESSNA 182F CESSNA 182G CESSNA 182H CESSNA 182J CESSNA 182K CESSNA 182L CESSNA 182M CESSNA 182N | Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-588 in Cessna 182E, 182F, 182G,182H, 182J,182K, 182L, 182M, and 182N, also when modified per STC SA1382WE (Robertson STOL) | | Bulletin No. 588, Rev. 6 or later approved revision,Master Drawing List No. 92787, Rev. F, dated 10-23-02, or later approved revisionAFM-Supplements as mentioned belowInstructions for Continued Airworthiness, P/N 81293, Rev. 3 or later approved version | FAA/DAS Approved Supplemental Flight Manual, Part No. 891174, Rev. 2, dated 6-11-01, or later approved revisions, is required for Cessna Models 182E, 182F, and 182G.FAA/DAS Approved Supplemental Flight Manual, Part No. 891175, Rev. 2, dated 6-11-01, (...) | 19.10.2006 |
| 10015927 | S-TEC CORPORATION | EASA.IM.A.S.02063 | REV. 1 | FAA STC SA7199SW-D | BEECH 100 BEECH A100 BEECH B100 | Installation of S-TEC Yaw Damper System, Model ST-302 | | Bulletin No. 402, Revision 1, dated 4/09/1990Master DrawingList No. 92496, Revision A, dated 4/09/1990 and/or later approved revision of the above data (28 Volt System) | 1. Also eligible on Models 100 and B1000 when modified per STC SA3823NM (Lower Aft Fuselage Strakes).2. Also eligible on Model A1000 when modified per STC SA3828NM (Lower Aft Fuselage Strakes) with or without STC SA16779SO (750 S.H.P. Engines).3. Fl (...) | 17.10.2006 |
| 10015928 | S-TEC CORPORATION | EASA.IM.A.S.02064 | REV. 1 | FAA STC SA7198W-D | BEECH 100 BEECH A100 BEECH B100 | Installation of S-TEC Systems 65 Two Axis Automatic Flight Guidance System, Model ST-301 | | Bulletin No. 401, Revision 4, dated 5-24-93Master Drawing List No. 92495, Revision D, dated 5-24-93 with Optional Flight Director/Steering Horizon or later approved revisions on the above data (28 Volt Systems). | 1. Also eligible on Models 100, A100 and B1000 when modified per STC SA3828NM (Lower Aft Fuselage Strakes).2. Also eligible on Model A100 when modified per STC SA1679SO (750 S.H.P. Engines).3. Airplane Flight Manual Supplement, P/N 89827S,dated 9-29 (...) | 17.10.2006 |
| 10015996 | S-TEC CORPORATION | EASA.IM.A.S.02152 | | FAA STC SA09124AC-D | PIPER PA-60-601P PIPER PA-60-602P | Installation of S-TEC 55/55X two axis automatic flight control system in Aerostar models | | - STC SA09124AC-D- AFM Suppl. P/N891204S Rev. 2- ICA P/N 81319 Rev. 2- S-TEC Bulletin 706 Rev. 2- MDL No. 92812 Rev. B | 1. According to AFMS, Section 2 Operating Limitation2. Instructions for continued airworthiness acc. Doc. P/N 81319 Rev. 23. Maintenance Instructions 2.-4. must be performed every 100 flight hours or once every calendar year, whichever comes first.4. This STC is approved only for the pro | 29.08.2006 |
| 10016052 | S-TEC CORPORATION | EASA.IM.A.S.02224 | | FAA STC SA09262AC-D | CESSNA 182Q CESSNA 182R CESSNA 182T | Installation of S-TEC System 20/30 Single and Two Axis Automatic Flight Guidance System, Model ST-718-20/30 in Cessna 182Q, 182R and T182, also when modified per STC SA1382WE (Robertson STOL) | | Bulletin No. 818, dated 30-Jul-1997 or later approved revisionMaster Drawing List No. 921016, dated 30-Jul-1997, or later approved revisionFAA/DAS Approved Supplemental Flight Manual Supplemental, P/N 891462, dated 04-Aug-1997, or later approved rev (...) | For 182Q the maximum aft c.g. is limited to 46.0 inches aft of datum when autopilot is installed.installer must check availability of EASA-approval for Robertson STOL STC No. when aeroplane is modified per that STC | 14.12.2006 |
| 10016053 | S-TEC CORPORATION | EASA.IM.A.S.02225 | | FAA STC SA09218AC-D | GRUMMAN GA-7 | Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-661 in Grumman American GA-7 | | Bulletin No. 761, Rev. 2, dated 19-Mar-2002 or later approved revision,Master Drawing List No. 92986, Rev. B, dated 19-Mar-02, or later approved revisionFAA/DAS Approved Pilots Operating Handbook and / or Airplane Flight Manual Supplemental, Part N (...) | -- | 14.12.2006 |
| 10016054 | S-TEC CORPORATION | EASA.IM.A.S.02226 | | FAA STC SA09316AC-D | CESSNA R182 CESSNA TR182 | Installation of S-TEC System 20/30 Single and Two Axis Automatic Flight Guidance System, Model ST-759-20/30 in Cessna R182, and TR182. | | - Bulletin No. 859, dated 15-Dec-1997 or later approved revision.- Master Drawing List No. 921055, dated 15-Dec-1997, or later approved revision.- FAA/DAS Approved Pilots Operating Handbook and / or Airplane Flight Manual Supplemental, P/N (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 18.12.2006 |
| 10016080 | S-TEC CORPORATION | EASA.IM.A.S.02264 | | FAA STC SA09201AC-D | CESSNA 172R CESSNA 172S | Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-563 in Cessna 172R and 172S | | Bulletin No. 663, Revision 5, dated 01-Feb-2001 or later approved revision,Master Drawing List No. 92732, Revision E, dated 01-Feb-2001, or later approved revisionFAA/DAS Approved Pilots Operating Handbook and / or Airplane Flight Manual Supplement, (...) | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 16.01.2007 |
| 10016103 | S-TEC CORPORATION | EASA.IM.A.S.02293 | | FAA STC SA09160AC-D | PIPER PA-32-260 PIPER PA-32-300 PIPER PA-32R-300 | Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-640 | | Bulletin No. 740, Rev. 2, dated 28-Jun-2001 or later approved revision,Master Drawing List No. 92944, Rev. B, dated 28-Jun-2001, or later approved revision,FAA/DAS Approved Pilots Operating Handbook, and / or Airplane Flight Manual Supplement, Part (...) | | 28.02.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------|-------------------|----------------|----------------------|---|---|---------------------|--|--|---------------|
| 10016146 | S-TEC CORPORATION | EASA.IM.A.S.02347 | | FAA STC SA09397CA-D | PIPER PA-31 PIPER PA-31-300 PIPER PA-31-325 PIPER PA-31-350 | Installation of S-TEC System 65 Two Axis Automatic Flight Guidance System with Autotrim, Model ST-849, and with Optional Flight Director/Steering Horizon. | | - Bulletin No. 949, Rev. 5, dated 01-Jul-2003 or later approved revision, - Master Drawing List No. 921136, Rev. E, dated 01-Jul-2003, with Optional Flight Director / Steering Horizon, or later approved revision-FAA/DAS Approved Pilot's Oper (...) | This STC is also eligible on: - PA-31 and PA-31-300 when modified per STC SA97050, - PA-31-325 when modified per STC SA000485E, - PA-31-350 when modified per STC SA115150 and / or any of STC SA5796NW, STC SA001925E, STC SA000395E, STC SA00 (...) | 23.02.2007 |
| 10016175 | S-TEC CORPORATION | EASA.IM.A.S.02403 | | FAA STC SA09139AC-D | PIPER PA-44-180 | Installation of S-TEC Manual Electric Trim System, Model ST-619, in above mentioned airplane Type / Variant | | Bulletin No. 719, Revision IR, dated 05-Mar-1996, or later approved revisionFor field installations: Master Drawing List No. 92898, Initial Revision, dated 05-Mar-1996, or later approved revisionsFor Piper Factory Installations: Master Drawing List (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.The (...) | 15.06.2007 |
| 10016194 | S-TEC CORPORATION | EASA.IM.A.S.02431 | | FAA STC SA8886SW-D | CESSNA 206 CESSNA P206 CESSNA P206A CESSNA P206D CESSNA P206E CESSNA TP206C CESSNA TP206D CESSNA TU206B CESSNA TU206C CESSNA TU206D CESSNA TU206E CESSNA TU206G CESSNA U206 CESSNA U206B CESSNA U206C CESSNA U206D CESSNA U206E CESSNA U206F CESSNA U206G | Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-556 in above mentioned airplaneTypes / Variants, also when modified per STC as listed in 3.5 (Limitations and Conditions) | | Bulletin No. 656, Revision 3, dated 08-Mar-2001, or later approved revisionMaster Drawing List No. 92722, Revision C, dated 08-Mar-2001, or later approved revision (28 Volt System)FAA/DAS Approved Pilot's Operating Handbook and/or AirplaneFlight Man (...) | This STC is also eligible on variants 206, P206, U206, P206A, TP206A, U206A, TU206A, P206B, TP206B, U206B, TU206B, P206C, TP206C, U206C, TU206C, P206D, TP206D, U206D, P206E, TP206E, U206E, TU206E, U206F, and TU206F when converted to 28 Volt (...) | 05.07.2007 |
| 10016196 | S-TEC CORPORATION | EASA.IM.A.S.02434 | | FAA STC SA09352AC-D | | Installation of S-TEC System, Model 20/30 Single and Two Axis Automatic Flight Guidance Systems, Model ST-810 | | FAA approved S-TEC Bulletin No.910, Revision 4, dated 26-Dec-2000, or later approved revisionFAA approved S-TEC Master Drawing List No. 921085, Revision D, dated 26-Dec-2000, or later approved revision.FAA/DAS Approved Pilot's Operating Handbook and (...) | This STC is eligible on variants M-5-210C, M-5-235C, M-6-235, MX-7-180, M-7-235, M-7-235A, M-7-235B, M-7-235C and MX-7-235 in landplane, floatplane, or amphibian configuration.This STC is eligible on variants MX-7-180A, MX-7-180B, MX-7-180C, MXT-7 (...) | 03.07.2007 |
| 10016203 | S-TEC CORPORATION | EASA.IM.A.S.02445 | | FAA STC SA09287AC-D | PIPER 31 AND SERIES PIPER PA-28R-180 PIPER PA-28R-200 PIPER PA-28R-201T | Installation of S-TEC System, Model 20/30 Single and Two Axis Automatic Flight Guidance Systems, Model ST-729 | | - S-TEC Bulletin No.829, dated 08-Oct-1997, Initial Issue, or later approved revision# S-TEC Master Drawing List No. 921031, dated 08-Oct-1997, Initial Issue or later approved revision# FAA/DAS Approved Pilot's Operating Handbook and/or Air (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 26.07.2007 |
| 10016278 | S-TEC CORPORATION | EASA.IM.A.S.02567 | | FAA STC SA09251AC-D | PIPER PA-28-161 PIPER PA-28-181 PIPER PA-28-236 | Installation of S-TEC System 20/30 Single and Two Axis Automatic Flight Guidance Systems, Model ST-703-20/30. | | FAA approved S-TEC Bulletin No. 803, Revision 0, dated 23-Jun-1997, or later approved revisionFAA approved Master Drawing List No. 921009,Revision 0, dated 23-Jun-1997, or later approved revisionFAA / DAS approved Pilot's Operating Handbook and / or (...) | This STC is also eligible on: - PA-28-151 when modified per STC SA29695W (180 HP Engine Conversion)If airplane is modified per the above mentioned STC, installer must check availability of EASA-approval. | 22.10.2007 |
| 10016317 | S-TEC CORPORATION | EASA.IM.A.S.02623 | REV. 1 | FAA STC SA09001AC-D | CESSNA 310I CESSNA 310J CESSNA 310K CESSNA 310L CESSNA 310N CESSNA 310P CESSNA 310Q CESSNA 310R CESSNA T310P CESSNA T310Q CESSNA T310R | Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-594, also when modified per FAASTC as listed in 3.5 (Limitations and Conditions). | | FAA approved Bulletin No. 694, Revision 4, dated 11-May-2001, or later approved revisionFAA approved Master Drawing List No. 92794, Revision D, dated 11-May-2001, or later approved revision (28 Volt System)FAA/DAS Approved Supplemental Flight Manual (...) | This STC is also eligible on variants 310K, 310L, and 310N when modified per STC SA41250 (300 H.P. Coleman Conversion).This STC is also eligible on variants T310P, and T310Q when modified per STC SA27505W (RAM Modification.Installer must check a (...) | 22.11.2007 |
| 10016405 | S-TEC CORPORATION | EASA.IM.A.S.02734 | | FAA STC SA09298AC-D | PIPER PA-31 PIPER PA-31-325 | Installation of S-TEC SYSTEM 55/55X Two Axis Automatic Flight Guidance System, Model ST-751, also when modified per FAASTC as listed in 3.5 (Limitations and Conditions). | | FAA approved Bulletin No. 851, Revision 3, dated 3-12-01 or later FAA approved revisions (28 Volt System)FAA approved Master Drawing List No. 921041, Rev. C, dated 3-12-01 or later FAA approved revisions (28 Volt System)FAA/DAS Approved Pilot's Opera (...) | This STC is also eligible on Models PA-31 when modified perSTC SA97050 (350 hp Coleman Conversion)This STC is also eligible on Model PA-31-325 when modified per STC SA000485E (Vortex Generators System).Prior to installation of this modification the (...) | 22.05.2008 |
| 10016425 | S-TEC CORPORATION | EASA.IM.A.S.02763 | | FAA STC SA09132AC-D | CESSNA 172M CESSNA 172N | Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-612 in Cessna 172M and 172N. | | Bulletin No. 712, Revision 2, dated 28-Feb-2001 or later approved revision.Master Drawing List No. 92827, Revision B, dated 28-Feb-2001, or later approved revision (14 Volt System).FAA/DAS Approved or Airplane Flight Manual Supplement, P/N891218 Rev (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 08.05.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------|---------------------|----------------|------------------------------|---|--|---|--|--|---------------|
| 10016442 | S-TEC CORPORATION | EASA.IM.A.S.02787 | | FAA STC SA09233AC-D | PIPER PA-28-161 PIPER PA-28-181 PIPER PA-28-236 | Installation of S-TEC System 30 ALT Altitude Hold System, Model ST 684 (28 Volt System) | | FAA approved S-TEC Bulletin No.784, dated 29-May-1997, Initial Issue, or later approved revisionFAA approved S-TEC Master Drawing List No. 92998, dated 29-May-1997, Initial Issue or later approved revisionFAA/DAS Approved Pilots#s OperatingHandbook a (...) | The approval holder shall fulfil the obligations of Part 21, Paragraph 21.A109 | 13.05.2008 |
| 10016604 | S-TEC CORPORATION | EASA.IM.A.S.03019 | | FAA STC SA09257AC-D | PIPER PA-28-140 PIPER PA-28-150 PIPER PA-28-160 PIPER PA-28-180 PIPER PA-28-235 | Installation of S-TEC System 30 ALT Altitude Hold System, Model ST-712 (14 Volt System) | | FAA approved S-TEC Bulletin No.812, dated 08-July-1997, Initial Issue, or later approved revisionFAA approved S-TEC Master Drawing List No. 921012, dated 08-July-1997, Initial Issue or later approved revisionFAA/DAS Approved Pilots Operating Handbo (...) | 1 Prior to installation of this modification the installer must verify the availability of EASA approval for one or more mentioned STC(s) when the product is modified per that/those STC(s).1 Prior to installation of this modificationthe instal (...) | 14.01.2009 |
| 10028097 | S-TEC CORPORATION | P-EASA.IM.A.S.03269 | | FAA STC SA5357SW-D | PA-28-151, PA-28-161, PA-28-181, PA-28-236, PA-28-201T | Installation of S-TEC Manual Electric Trim System, Model ST-156 (14 VoltSystem) | The Certification Basis for the original product remains applicable tothis certificate/ approvalThe certified noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval | Applicable revisions:-FAA Approved S-TEC Bulletin No. 256, dated 21-Nov-84, Initial Issue, orlater revision-FAA Approved S-TEC Master Drawing List No. 92306, dated 21-Nov-84,Initial Issue, or later revision-FAA Approved Supplement to Pilot's Operati (...) | Compatibility of this modification with other previously approvedmodifications must be determined by the installer.The Approval of the installation of this system is not applicable tospecific aircraft having other previously approved modifications, (...) | 01.12.2009 |
| 10028331 | S-TEC CORPORATION | | | FAA STC SA09401AC-D (ST-853) | 401, 401A, 401B 402A, 402B, 402C | Installation of S-TEC System 55/55X Two Axis Automatic Flight GuidanceSystem, Model ST-853 (28 Volt System) | The Certification Basis for the original product remains applicable tothis certificate/ approval | Applicable revisions-FAA Approved S-TEC Bulletin No. 953, dated 08-Mar-99, Initial Issue, orlater revision-FAA ApprovedS-TEC Master Drawing List No. 921151, dated 08-Mar-99,Initial Issue, or later revision-FAA/DAS Approved Pilot's Operating Handbook (...) | Compatibility of this modification with other previously approvedmodifications must be determined by the installer.The Approval of the installation of this system is not applicable tospecific aircraft having other previously approved modifications, (...) | 17.12.2009 |
| 10029937 | S-TEC CORPORATION | | | FAA STC SA09140AC-D | 210, 210A, 210B, 210C | Installation of S-TEC System 55/55X Two Axis Automatic Flight GuidanceSystem, Model ST-621 (14 Volt System) | The Certification Basis for the original product remains applicable tothis certificate/ approvalThe certified noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval | -FAA Approved S-TEC Bulletin No. 721, dated 18-July-01, Revision 3, orlater revision-FAA Approved S-TEC Master Drawing List No. 92899, dated 18-July-01,Revision C, or later revision-FAA Approved Supplemental Flight Manual, P/N 891234, Revision 2, da (...) | None | 04.05.2010 |
| 10032286 | S-TEC CORPORATION | | | FAA STC SA8910SW-D | BEECH S35, V35, V35A AND V35B | Installation of S-TEC System 55/55X Two Axis Automatic Flight GuidanceSystem, Model ST-577 (14 Volt System) | The Certification Basis for the original product remains applicable tothis certificate/ approvalThe certified noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval | FAA-Approved S-TEC Bulletin No. 677, dated 04-Dec-09, Revision 6, orlater approved revisionFAA-Approved S-TEC Master Drawing List No. 92756, dated 04-Dec-09,Revision F, or later approved revisionFAA-Approved Pilot's Operating Handbook and/or Airplan (...) | This STC is also eligible on Beech models S35, V35, V35A, and V35B whenmodified by STC SA1035WE (turbocharged engine) or STC SA523NM (turbonormalizing system) and any combinationof the following STCs: STCSA09074SC (vortex generators), STC SA4-1629 (tip tanks). | 22.10.2010 |
| 10032336 | S-TEC CORPORATION | | | FAA STC SA09320AC-D | CESSNA F182P, F182Q | Installation of S-TEC System 20/30 Single and Two Axis Automatic FlightGuidance Systems, Model ST-762-20/30 (14 Volt System) | The Certification Basis for the original product remains applicable tothis certificate/ approvalThe certified noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval | FAA-approved S-TEC Bulletin No. 862, dated 29-May-09, Revision 6FAA-approved S-TEC Master Drawing List No. 921057, dated 10-Dec-09,Revision Gfor S-TEC System 20: FAA-approved Pilots' Operating Handbook and/orAirplane Flight Manual Supplement, P/N 89 (...) | This STC is also eligible on Cessna models F182P and F182Q when modifiedby: 1. STC SA485SW (Wren conversion) and STC SA3825SW (260 H.P. engineconversion) or 2. STC SA950CE (HortonSTOL) and/or STC SA2285CE (Hortonlap gap seals) and STC SA3825SW (260 (...) | 28.10.2010 |
| 10032443 | S-TEC CORPORATION | | | FAA STC SA7818SW-D | 35-C33, F33C | Installation of S-TEC System 40/50 Single and Two Axis Automatic FlightGuidance Systems, Model ST-492-40/50 (28 Volt System) | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Installation in accordance withFAA Approved S-TEC Bulletin No. 592, dated 18-June-92.FAA Approved S-TEC Master Drawing List No. 92565, dated 18-June-92.Operation in accordance withFAA Approved Airplane Flight Manual Supplement, P/N 89975, dated26-Ju (...) | none | 08.11.2010 |
| 10032444 | S-TEC CORPORATION | | | FAA STC SA7819SW-D | 35-C33, F33C | Installation of S-TEC Manual Electric Trim System, Model ST-493 (28 VoltSystem) | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Installation in accordance withFAA Approved S-TEC Bulletin No. 593, dated 18-June-92FAA Approved S-TEC Master Drawing List No. 92566, dated 18-June-92.Operation in accordance withFAA Approved Airplane Flight Manual Supplement, P/N 89977, dated26-June (...) | none | 08.11.2010 |
| 10033463 | S-TEC CORPORATION | | | FAA STC SA8401SW-D | TB9, TB10, TB20, TB21, TB200 | Installation of S-TEC System 55/55X Two Axis Automatic Flight GuidanceSystem, Model ST-547 (28 Volt System) | The Certification Basis for the original product remains applicable tothis certificate/ approval. | 1) Installation in accordance with both:- FAA Approved S-TEC Bulletin No. 647, dated 02-Nov-09, Revision 4- FAA Approved S-TEC Master Drawing List No. 92697, dated 02-Nov-09,Revision D2) Operation in accordance with:a) For Socata models TB9, TB 10, (...) | Optional Autotrim System not eligible on Socata model TB 9, TB 10, andTB 200 (Group Aerospatiale) | 24.01.2011 |
| 10033582 | S-TEC CORPORATION | | | FAA STC SA5210SW-D | H35, J35, K35, M35, N35, P35 | Installation of S-TEC 40/50 Single and Two Axis Automatic FlightGuidance Systems, Model ST-188-40/50 (14V - System) | Design:Non-significant changeThe Certification Basis of theoriginal product remains applicable tothis changeEnvironmental Protection:The certified noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval | FAA approved S-TEC Bulletin No.288, dated 20-Jul-1983, Initial Issue, orlater approved revisionFAA approved S-TEC Master Drawing List No. 92216, dated 20-Jul-1983,Initial Issue, or later approved revisionFAA Approved Supplement to Pilots#s Operating (...) | None | 28.01.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------|--------------|----------------|--------------------------|---|--|--|--|--|---------------|
| 1003820 | S-TEC CORPORATION | | | FAA STC SA09285AC-D | PA-31-350 PA-31, PA-31-300, PA-31-325 | Installation of S-TEC 20/30 Single and Two Axis Automatic Flight Guidance Systems, Model ST-727-20/30 (28V - System) | The Certification Basis of the original product remains applicable to this change. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA approved S-TEC Bulletin No.827, Revision 4, dated 26-Apr-2001, or later approved revision FAA approved S-TEC Master Drawing List No. 921029, Revision D, dated 26-Apr-2001, later approved revision FAA / DAS Approved Pilot's Operating Handbook and/or (...) | This STC is also eligible on models PA-31 and PA-31-300 when modified per FAA STC SA970SO (350 H.P. Coleman Conversion) This STC is also eligible on model PA-31-325 when modified per FAA STCSA00048SE (Vortex Generator System) This STC is also eligible (...) | 10.02.2011 |
| 10038328 | S-TEC CORPORATION | | | FAA STC SA8392SW-D | PA-32-301, PA-32-301T PA-32R-301, PA-32R-301T | Installation of S-TEC system 55/55X Two Axis Automatic Flight Guidance Systems, Model ST-534 (14V - System). | The Certification Basis of the original product remains applicable to this change. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA approved S-TEC Bulletin No.634, Revision 4, dated 16-Mar-2011, or later FAA approved revision FAA approved S-TEC Master Drawing List No. 92683, Revision D, dated 16-Mar-2011, or later FAA approved revision FAA / DAS Approved Pilot's Operating Handbook and/or (...) | Installer must check availability of EASA-approval for one or more referenced STC when airplane is modified per that / those STC(s) | 07.02.2012 |
| 10038425 | S-TEC CORPORATION | | | FAA STC SA09307AC-D | 401/401A/401B/402A/402B/402C | Installation of S-TEC system 20/30 single and two axis Automatic Flight Guidance Systems. Installation Model ST-743 (28V- System) according to bulletin N. 843. | The Certification Basis of the original product remains applicable to this change. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA approved S-TEC Bulletin No.843, dated 11-Oct-1997, or later FAA approved revision FAA approved S-TEC Master Drawing List No. 921046 dated 11-Oct-1997, or later FAA approved revision FAA / DAS Approved Pilot's Operating Handbook and/or Airplane FL (...) | Installer must check availability of EASA-approval for one or more referenced STC when airplane is modified per that / those STC(s) | 15.02.2012 |
| 10039195 | S-TEC CORPORATION | | REV. 1 | FAA STC SR02344LA | BELL 407 | Installation of HelisAS stability augmentation system and autopilot. Validation of FAA STC SR02344LA | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. FAR 27, with following amendments and related paragraphs: Amdt 11: 1322; Amdt 13: 1321, (...) | Master Data List, HelisAS-MDL-407 revision D dated March 19, 2012-Rotorcraft Flight Manual Supplement, HelisAS-RFMS-407 revision N/C dated March 1, 2011-Instructions for continued Airworthiness (ICA), Document No. AA-09109 Revision B dated November 2 (...) | VFR only | 18.04.2012 |
| 10039237 | S-TEC CORPORATION | | | FAA STC NO. SA09350 AC-D | BEECH 95/B95/B95A/D95A/E95 | Installation of S-TEC system 20/30 single and two axis Automatic Flight Guidance Systems, model ST-908-20/30, according to bulletin N. 908 rev. 1 dated 01-Dec-2008 and with Master Drawing List No. 921083, Rev. A, dated 01-Dec-2008 or later FAA approved revision (28V - System). | The Certification Basis of the original product remains applicable to this change. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA approved S-TEC Bulletin N. 908, rev.1 dated 01-Dec-2008, or later FAA approved revision FAA approved S-TEC Master Drawing List No. 921083, Rev. A, dated 01-Dec-2008, or later FAA approved revision FAA / DAS Approved Pilot's Operating Handbook and/or (...) | Installer must check availability of EASA-approval for one or more referenced STC when airplane is modified per that / those STC(s). | |
| 10039264 | S-TEC CORPORATION | | REV. 1 | FAA STC SA09010AC-D | SEE EASA APPROVED MODEL LIST | Installation of S-TEC system 55/55X two axis Automatic Flight Guidance Systems, model ST-589, according to bulletin N. 689 rev.4 dated 11-Jan-2010 and with Master Drawing List N. 92790, Rev. D, dated 11-Jan-2010 or later FAA approved revision (14V - System). | The Certification Basis of the original product remains applicable to this change. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA approved S-TEC Bulletin N. 689 rev.4, dated 11-Jan-2010 or later FAA approved revision FAA approved S-TEC Master Drawing List No. 92790, Rev. D, dated 11-Jan-2010, or later FAA approved revision FAA / DAS Approved Pilot's Operating Handbook and/or (...) | Installer must check availability of EASA-approval for one or more referenced STC when airplane is modified per that / those STC(s). | 25.04.2012 |
| 10039317 | S-TEC CORPORATION | | | FAA STC SA8364SW-D | 401, 401A, 401B 402A, 402B, 402C | Installation of S-TEC system 40/50 single and two axis Automatic Flight Guidance Systems. Installation Model ST-505-40/50 (28V - System) according to bulletin N. 605 rev.6 dated 21-Jan-2010 and master drawing list n. 92582 rev. F, dated 21-Jan-2010 or later FAA approved revision of above documents | The Certification Basis of the original product remains applicable to this change. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA approved S-TEC Bulletin No.505 Rev. 6 dated 21-Jan-2010, or later FAA approved revision FAA approved S-TEC Master Drawing List n. 92582 Rev. F dated 21-Jan-2010, or later FAA approved revision FAA / DAS Approved Pilot's Operating Handbook and/or A (...) | Installer must check availability of EASA-approval for one or more referenced STC when airplane is modified per that / those STC(s) | 25.04.2012 |
| 10039903 | S-TEC CORPORATION | | | FAA STC SA09540 AC-D | A99A, B99, C99 | Installation of S-TEC system 30 two axis Automatic Flight Guidance Systems (ST-923), in accordance with Master Data List No. ST-923-L-0001, Revision A, dated 24-Aug-06 or later FAA approved revision (28V - System). | The Certification Basis of the original product remains applicable to this change. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA approved S-TEC Master Data List No. ST-923-L-0001, Revision A, dated 24-Aug-06 or later FAA approved revision FAA / DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement No. ST-923-M-0001 Revision-, dated 23-Aug-2006 (...) | Installer must check availability of EASA-approval for one or more referenced STC when airplane is modified per that / those STC(s). If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. | 31.05.2012 |
| 10040585 | S-TEC CORPORATION | | | FAA STC SR002346LA | BELL 206B, 206L, 206L-1, BELL 206L-3, 206L-4 | Installation of HelisAS stability augmentation system and autopilot. Validation of FAA STC SR02346LA | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. FAR 27, with following amendments and related paragraphs: Amdt 11: 1322; Amdt 13: 1321, (...) | Rotorcraft Flight Manual Supplement, HelisAS-RFMS-206 revision N/C dated March 1, 2011- Master Data List, HelisAS-MDL-206B revision E dated March 19, 2012- Instructions for Continued Airworthiness (ICA), Document No. AA-09021 Revision B dated Novemb (...) | VFR only | 17.07.2012 |
| 10040647 | S-TEC CORPORATION | | REV. 1 | FAA STC NO. SA7178SW-D | T210K T210K | Installation of S-TEC manual Electric trim system, (ST-433), according to Bulletin n. 533, dated 11-May-1988 and Master Drawing List No. 92474, dated 11-May-1988 or later FAA approved revision (14V - System). | Design: Non-significant change The Certification Basis of the original product remains applicable to this change. Environmental Protection: The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA approved S-TEC Master Drawing List No. 92474 dated 11-May-1988 or later FAA approved revision FAA / DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement No. 89784 dated 18 May-1988 or later FAA approved revision. | T210K is eligible also when modified per STC SA2689SW (RAM modification). Installer must check availability of EASA-approval for one or more referenced STC when airplane is modified per that / those STC(s). If the holder agrees to permit another person (...) | 17.07.2012 |
| 10041971 | S-TEC CORPORATION | | | FAA STC SR02345LA | A5350B A5350B1 A5350B2 A5350B3 A5350BA A5350D EC130B4 | Installation of HelisAS stability augmentation system and autopilot. Validation of FAA STC SR02345LA. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. FAR 27, with following amendments and related paragraphs: Amdt 11: 1322; Amdt 13: 1321, (...) | -, Rotorcraft Flight Manual Supplement, HelisAS-RFMS-A5350/EC130 revision N/C dated May 10, 2011; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)- (...) | VFR only. | 29.10.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|--------------------|----------------|------------------------|---|--|---|--|--|---------------|
| 10042013 | S-TEC CORPORATION | | | FAA STC NO. SA5301SW-D | 58 58A | Installation of S-TEC system 40/50, model ST 224-40/50, according to bulletin n. 324 dated 22-Sep-1983 and Master Drawing List No. 92252, dated 22-Sep-1983 (28V - System). | Design: Non-significant change The Certification Basis of original product remains applicable to this change. Environmental Protection: The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA approved S-TEC Master Drawing List No. 92252, dated 22-Sep-1983 (28V-System); FAA / DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement No. 89350 dated 5-Apr-1984 for system 40; FAA / DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement No. 89351 dated 5-Apr-19 (...) | FAA / DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement No. 89350 dated 5-Apr-1984 is required for system 40; FAA / DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement No. 89351 dated 5-Apr-19 (...) | 31.10.2012 |
| 10042477 | S-TEC CORPORATION | | | | BN-2A-2, BN-2A-8 BN-2A-20, BN-2A-26 BN-2A-6 BN-2B-20, BN-2B-26 | Installation of S-TEC System 55/55X Two-Axis Automatic Flight Guidance System Model ST572. | | - Bulletin No 672 Revision 2 dated August 21, 2002 & M.D.L. No 92744 Revision B dated August 21, 2002 or later approved revision-AFM Supplement EASA891140 dated May 05, 2005 or later EASA approved revision | Only applicable to Model BN-2A-6 when modification NB/M430 and NB/M631 are installed. | 31.05.2005 |
| 10042955 | S-TEC CORPORATION | EASA, IM.A.S.01152 | | | CESSNA 180, 180A, 180B, 180C CESSNA 180D, 180E, 180F | Installation of S-TEC System 40/50 Single & Two Axis automatic flight guidance systems, Model ST-196-40/50 | | Bulletin No. 296 Revision 8 dated March 22, 1999. Master Drawing List No. 92232 Revision H dated March 22, 1999 or later EASA approved Revision (14 Volt System), S-TEC AFM Supplement P/N 89484 applicable to system 40 dated July, 191984 or later EASA appr (...) | 1. This STC applies to Cessna Models 180, 180A, 180B, 180C, 180D, 180E, 180F (Floatplane or amphibian) & the above models when equipped with FAASTC SA1435WE (Robertson STOL) or FAA STC SA967CE (Horton STOL) and / or FAA STC SA1705WE (Internal tanks). (...) | 18.07.2005 |
| 10042980 | S-TEC CORPORATION | EASA, IM.A.S.01629 | | FAA STC SA8400SW-D | TB20, TB21 | Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-546 | | Bulletin No 546 Revision 3, dated 23 August 2001 and Master drawing list No 92695, Revision C, dated 23 August 2001 or later approved revisions of the above data (14 Volt system) EASA approved Pilot's Operating Handbook and/or Airplane Flight Manual Suppl (...) | 1. The STC is issued as result of the validation of the S-TEC Corporation STC No SA8400SW-D approved by the FAA on August 23, 2001. This STC is approved only for the product configuration as defined in the approved design data referred to in the par (...) | 12.12.2005 |
| 10044442 | S-TEC CORPORATION | | | FAA STC SA5372SW-D | AA-5A, AA-5B | Installation of S-TEC system 60 Two Axis Autopilot model ST-278, according to bulletin N.378 rev.1 dated 3-Oct-1986 (FAASTC SA5372SW-D). | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | According to S-TEC Master Drawing List N. 92313 rev. A dated 3-Oct-1986 with optional Flight Director/Steering Horizon. To be operated in accordance with S-TEC AFM-S P/N 89588 dated 14-Feb-1985. To be operated in accordance with S-TEC AFM-S P/N 89588-1 (...) | The installation of this modification by third persons is subject to written permission of the approval holder and disposal of the approved appropriate documentation. Prior to installation of this modification it must be determined that the interrelation (...) | 12.04.2013 |
| 10044510 | S-TEC CORPORATION | | | FAA STC SA08998AC-D | 36, A36, F33A | Installation of S-TEC system 55/55X Two Axis Autopilot model ST-582, according to bulletin N.682 rev.4 dated 15-Jan-2002 (FAASTC SA08998AC-D). | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -, According to S-TEC Master Drawing List N. 92761 rev.D dated 15-Jan-2002; -, To be operated in accordance with S-TEC AFM-S P/N 891156 dated 13-Apr-1995 and AFM-S P/N 891156 rev.1 dated 20-Mar-2001 and AFM-S P/N 891156 rev.2 dated 18-Jan-2002 in respect (...) | -, For Beech model F33A, eligibility is limited to S/N CE-316 and up; also, eligible to models 36, A36 and F33A S/N GE-316 and up when modified per STC SA153EA-. Also eligible to models 36, A36 when modified per STC SA5223NM-, AFMS P/N 891156 dat (...) | 17.04.2013 |
| 10044926 | S-TEC CORPORATION | | | FAA STC SA09521AC-D | PA-46-500TP | Installation of S-TEC digital Flight control System (autopilot) Part No.01298-01-01-001 or 01304-01-01-001 1500 Autopilot with Autotrim and Yaw Dampner System, per report 97-13001-003, rev.2 dated 23-Sep-2003 and in accordance with ST-891 MDL No. 921213, rev.D dated 24-Jul-2009 | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment. Any paragraph indicated in the original FAA STC SA09521AC-D dated 10-Dec- (...) | - According to S-TEC Master Drawing List N. 921213 rev.D dated 24-Jul-2009; - To be operated in accordance with S-TEC AFM-S P/N 892112, dated 21-May-2002 or S-TEC AFM-S P/N 892112, rev.1 dated 01-Oct-2003 or S-TEC AFM-S P/N 892112, rev.2 dated 10-Dec-200 (...) | - S-TEC AFM-S P/N 892112, dated 21-May-2002 is required for Piper model PA-46-500TP for Magic 1500 Autopilot; - S-TEC AFM-S P/N 892112, rev.1 dated 01-Oct-2003 is required for Piper model PA-46-500TP for S-TEC 1500 Autopilot; - S-TEC AFM-S P/N 892112, (...) | 21.05.2013 |
| 10014400 | STEINBEIS FLUGZEUG- UND LEICHTBAU G | EASA.A.S.01695 | | | DG LS8-T | LS8-th Hydraulic extension system Exchange of the electric driven engine extension system to a hand-driven hydraulic system | | Change of pages in the Flight and Maintenance Handbook | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 06.07.2006 |
| 10034286 | STEINBEIS FLUGZEUG- UND LEICHTBAU G | P-EASA.A.S.03832 | | | LS 6 & LS 6-A | STC: "Winglet Retrofit" / "Nachrüstung mit Winglets" | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: JAR-22.375 from amendment 22/90/1 (for the winglets) of the JAE 22Change 4 | STC "Winglet Retrofit" / "Nachrüstung mit Winglets" EASA.A.S.03832- Supplement to the flight manual of the sailplane LS6 / LS6-a with retrofitted winglets "LS6-w"- Ergänzung des Flughandbuchs für das Segelflugzeug LS6 / LS6-a mit nachgerüsteten Wing (...) | None | 23.03.2011 |
| 10044570 | STERLING AVIATION TECHNOLOGIES Inc. | | | PMA APPROVAL PQ2158NM | PA-38-112 | Approval of PMA P/N SA12001 (replacing P/N 7753-05) - Vertical fin/Attach plate between Vertical Fin Forward Spar to a Fuselage Bulkhead (PMA approval PQ2158NM - as per FAA PMA letter dated July 27 2005). | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | PMA approval PQ2158NM - as per FAA PMA letter dated July 27 2005; Drawing n. SA12001. To be maintained according to ICA valid for P/N 7753-05 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/C (...) | Lifetime limit of this part is 5000h as detailed in the approved FAA documentation shipped with the part. The approval holder shall fulfill the obligations of Part 21, Point 21.A.118 Prior to installation of this design change it must be determined that (...) | 19.04.2013 |
| 10015396 | STG AEROSPACE Inc. | EASA, IM.A.S.01098 | | ST02993AT | BOEING 747-400 | Installation of SaTGlo Inc. Photoluminescent floor proximity emergency escape path marking system (FPPEPMS) in accordance with SaTGlo MDL Report M39-2, rev. B dated October 11, 2004 or later FAA-approved revision | | AFM Supplement SaTGlo Report No M39-11 rev. IR dated September 03, 2004 or later FAA-approved revision. Flight Attendant Manual (FAM) Supplement SaTGlo report No M39-13 rev. A dated October 7, 2004 or later FAA-approved revision. Flight Crew Operati (...) | 1. The photoluminescent FPPEPMS is only eligible for installation on aircraft model numbers with interior arrangements approved in the SaTGlo MDL, incorporating SaTGlo drawing SG-8747 rev B dated October 7, 2004 or later FAA-approved revision. 2 (...) | 16.03.2005 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------|-------------------|----------------|----------------------|--|---|---|--|---|---------------|
| 10015428 | STG AEROSPACE Inc. | EASA.IM.A.S.01161 | | FAA STC ST02691AT | AIRBUS A300 MODEL B2 AIRBUS A300 MODEL B4 AIRBUS A310-200 AIRBUS A310-300 | Installation of SaTGlo photoluminescent floor proximity emergency escape path marking system (FPPEPMS). - Validation of FAA STC No. ST02691AT). | | Master Drawing List No. L66-2 Revision D, dated 16 March 2004 or later FAA approved revision. Aircraft Flight Manual Supplement Document No. EASA L66-11, dated 25 August 2005, or later FAA approved revision. | 1. The approval of this STC is limited to the identified aircraft models with interior arrangements approved in the SaTGlo MDL No. L66-2 Revision D, dated 16 MAR 2004, for Airbus A300 incorporating SaTGlo Drawings SGA300 and for A310 incorporating (...) | 06.09.2005 |
| 10015429 | STG AEROSPACE Inc. | EASA.IM.A.S.01163 | | FAA STC ST02664AT | AIRBUS A340-200 AIRBUS A340-300 | Installation of SaTGlo photoluminescent floor proximity emergency escape path marking system (FPPEPMS) # Validation of FAA STC N° ST02664AT | | - Master Drawing List n° L69-2 Revision A, dated 30 January 2003, or later FAA approved revision.- Aircraft Flight Manual Supplement, Document N° EASA L69-11 Revision A, dated 07 September 2005, or later FAA approved revision. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. 2. The ap (...) | 16.11.2005 |
| 10015430 | STG AEROSPACE Inc. | EASA.IM.A.S.01164 | REV. 1 | FAA STC ST02529AT | ATR 42-200/300/320/500 ATR 72-101/102/201/202/211 ATR 72-212/212A | Installation of SaTGlo Photoluminescent Floor Proximity Emergency/Escapes Path Marking System (PFPEEPMS). (Validation of FAA STC ST02529AT)The STC installs a photoluminescent aisle path marking system (PFPEEPMS). This revision 01 adds ATR 72 aircraft models in the STC applicability. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -SaTGlo Master Drawing List, ref. P12-2 Revision C, dated 23 April 2007-EASA Aircraft Flight Manual Supplement, ref. EASA P12-11, dated 14 April 2011-Instruction for Continued Airworthiness, ref. P12-10 Revision C, dated 23 April 2007 or later revision (...) | None | 15.04.2011 |
| 10015431 | STG AEROSPACE Inc. | EASA.IM.A.S.01165 | | FAA STC ST02637AT | BOEING 717-200 | Installation of SaTGlo Inc. Phtoluminescent floor proximity emergency escape path marking system (FPPEPMS) | | | Cabin and exit lighting levels must be measured for acceptability on each installation in accordance with STG Aerospace Report No. P51-12, Revision A dated November 13, 2002 and P51-10, Revision I.R., dated August 15, 2002 or later approved revisions | 26.08.2005 |
| 10015432 | STG AEROSPACE Inc. | EASA.IM.A.S.01166 | | FAA STC ST01641AT | BOEING 727-100 SERIES BOEING 727-200 SERIES | Installation of a phtoluminescent Floor Proximity Emergency/Escapes Path Marking System according to Diversified Aviation Services Inc FAA STC No. ST01641AT. | | STG Aerospace Airplane Flight Manual Supplement Photoluminescent Aisle Path Marking System Boeing 727-100/200 Series, Doc No. EASAC51-11 Revision IR | | 21.09.2005 |
| 10015433 | STG AEROSPACE Inc. | EASA.IM.A.S.01168 | | FAA STC ST02221AT | BOEING 757-200 BOEING 757-300 | Installation of SaTGlo Inc. Phtoluminescent floor proximity emergency escape path marking system (FPPEPMS) | | SaTGLO MDL Report Number L21-2 Revision IR dated August 22, 2000 or later approved revision-AFM Supplement SaTGLO Report No L21-11 IR dated October 06, 2000 | Cabin and exit lighting levels must be measured for acceptability on each installation in accordance with STG Aerospace Report No. L21-12, Revision I.R, dated September 22, 2000 an L21-10, Revision I.R., dated September 21, 2000 or later approved revisions | 26.07.2005 |
| 10015434 | STG AEROSPACE Inc. | EASA.IM.A.S.01169 | | FAA STC ST02521AT | BOEING 767-200 SERIES BOEING 767-300 SERIES | Installation of SaTGlo Inc. Phtoluminescent floor proximity emergency escape path marking system (FPPEPMS) | | -SaTGLO MDL Report Number M40-2 Revision C dated May 16, 2002 or later approved revision-AFM Supplement SaTGLO Report No M40-11 Rev IR dated May 21, 2002 | Cabin and exit lighting levels must be measured for acceptability on each installation in accordance with STG Aerospace Report No. M40-12, Revision A dated October 11, 2001 and M40-10, Revision C, dated May 6, 2002 or later approved revisions | 26.07.2005 |
| 10015435 | STG AEROSPACE Inc. | EASA.IM.A.S.01170 | | FAA STC ST02687AT | BOMBARDIER CL-600-2B19 BOMBARDIER CL-600-2C10 | Installation of SaTGlo Inc. Phtoluminescent floor proximity emergency escape path marking system (FPPEPMS) | | -STG Aerospace MDL Report Number M41-2 revision D dated November 3, 2004 or later approved revision-AFM supplement No M41-11, Revision A dated June 24, 2004 or later approved revision | Cabin and exit lighting levels must be measured for acceptability on each installation in accordance with STG Aerospace Report No. M41-12, Revision A dated June 30, 2002 and M41-10, Revision I.R, dated June 5, 2003 or later approved revisions. | 28.07.2005 |
| 10015437 | STG AEROSPACE Inc. | EASA.IM.A.S.01175 | | FAA STC ST02131AT | DC 10-30 CF SERIES DC 10-30 SERIES DC-10-10 SERIES MD-11 | Installation of SaTGlo Inc. Photo luminescent floor proximity emergency escape path marking system in accordance with FAA STC ST02131AT | | STG Aerospace Revision record SG-DC10 dated 2/18/00 and report No's: K17-14 of Feb. 3, 2003, rev. I.R., K17-12 of Jan. 2, 2001, rev. B; K17-13 of Feb. 3, rev. B; K17-2 of Feb. 7, 2003, rev. C; K17-10 of Feb. 3, 2003, rev. B; K17-11 of Jan. 19, 2001 | 1. The Photoluminescent FPPEPM is only eligible for installation on aircraft model numbers with interior arrangements approved in the STG Aerospace document H61-2 Revision C, dated February 7, 2003, incorporating STG Aerospace drawing SG-DC10 rev B IS (...) | 15.11.2005 |
| 10015438 | STG AEROSPACE Inc. | EASA.IM.A.S.01176 | | FAA STC ST01893AT | BOMBARDIER DHC-7-100 BOMBARDIER DHC-7-102 | Installation of SaTGlo Inc. Phtoluminescent floor proximity emergency escape path marking system (FPPEPMS) | | -DAS MDL Report Number G34-2 Revision A dated April 18 19990 or later approved revisions-AFM Supplement DAS Report No G34-11, dated April 26, 1999 or later approved revisions | Cabin and exit lighting levels must be measured for acceptability on each installation in accordance with DAS Report No. G34-12, Revision A dated April 16, 1999 and G34-10, Revision A, dated April 17, 1999 or later approved revisions. | 28.07.2005 |
| 10015439 | STG AEROSPACE Inc. | EASA.IM.A.S.01179 | | FAA STC ST02406AT | FOKKER F.28 MARK 0070 FOKKER F.28 MARK 0100 FOKKER F.28 MARK 1000 FOKKER F.28 MARK 2000 FOKKER F.28 MARK 3000 FOKKER F.28 MARK 4000 | Installation of photo luminescent floor proximity emergency escape path marking system FAA STC ST02406AT | | STG Aerospace Revision record SG-F28 dated 7/9/99 and report no's: H61-12 of September 15, 2004, rev. C; H61-14 of June 28, rev. B, 2004; H61-13 of June 28, 2004 rev. E; H61-10 of August 20, 2004 rev. E. | 1. The Photoluminescent FPPEPM is only eligible for installation on aircraft model numbers with interior arrangements approved in the STG Aerospace document H61-2 Revision F, dated October 4, 2004, incorporating STG Aerospace drawing SG-F28rev D, date (...) | 15.11.2005 |
| 10016295 | STG AEROSPACE Inc. | EASA.IM.A.S.02592 | REV. 2 | FAA STC ST03465AT | B737-300/400/500 B737-700/800/900 | Installation of an STG Aerospace PowerSmart Wireless Emergency Primary Power System (WEPPS) in accordance with STG Master Drawing List (MDL) No. 71430000 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. The certification basis was updated in accordance with FAA policy for significant change (...) | -Master Drawing List (MDL) STG Doc. No. 71430000 Rev. G dated October 5 (<-> ->) 2009-Instructions for Continued Airworthiness (ICA) STG Doc. No. 7146006 Rev. B dated August 5, 2009-Enhanced Zonal Analysis Procedure (EZAP) Ref. 0710436002 Rev. IR-FAA (...) | None | 08.08.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|---------------------|----------------|----------------------|---|---|---|---|---|---------------|
| 10016490 | STG AEROSPACE Inc. | EASA.IM.A.S.02846 | | FAA STC ST01829AT | BOEING 737-100 BOEING 737-200 BOEING 737-300 BOEING 737-400 BOEING 737-500 BOEING 737-600 BOEING 737-700 BOEING 737-800 BOEING 737-900 | Installation of the SAFTGLO Non-electrical Photoluminescent Escape Path Marking System to include all 737 models. | | FAA STC ST01829AT, DAS/ SaTGlo MDL Report Number E74-2, Revision I.R., dated 9 February 1999, SaTGlo MDL Report Number E74-2, revision G, dated 13 June 2002* for installations occurring after 1 July 2002* or any later revision of the a (...) | The photo luminescent Floor Proximity Emergency Escape Path Marking System is only eligible for installation on aircraft model numbers with interior arrangements approved in the SaTGlo MDL No. E74-2, Rev. I.R. for the B 737 incorporating SaTGlo Dra (...) | 14.04.2008 |
| 10028318 | STG AEROSPACE Inc. | | | FAA TCDS T0001SE | B777-200 | Installation of SaTGlo, Inc photoluminescent floor proximity emergency escape path marking system | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC ST03684ATSTG report AC37-2 Master Drawing List rev.DAFM supplement AC37-11 dated May 5, 2009-12-16 Instruction for Continued Airworthiness AC37-20 dated March 16, 2009 or later revisions of the above listed documents approved by EASA in accordance with (...) | None | 17.12.2009 |
| 10028633 | STG AEROSPACE Inc. | P-EASA.IM.A.S.03118 | REV. 2 | FAA STC ST03629AT | B767-200, -300 | Installation of STG Aerospace PowerSmart Emergency Primary Power System (WEPPS) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this approval. | Master Drawing List 71670000 rev. D dated 29 October 2009 Instruction for Continued Airworthiness 71676006 rev. D dated 29 October 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 03.03.2011 |
| 10028634 | STG AEROSPACE Inc. | | REV. 1 | FAA STC ST03692AT | B757-200 | Installation of STG Aerospace PowerSmart Emergency Primary Power System (WEPPS) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this approval. | Master Drawing List 71440000 rev. C-Instruction for Continued Airworthiness 71446006 rev. B or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None | 23.03.2011 |
| 10031520 | STG AEROSPACE Inc. | | REV. 1 | FAA STC ST02311AT | DHC-8-102, -103, -106, -201 DHC-8-202, -301, -302, -401 DHC-8-402 | Installation of SaTGlo Inc. Photoluminescent floor proximity Escape Path Marking System (PFEPPMS) | FAR 25 as per EASA.IM.A.191 for DHC-8-100/200/300 Series/JAR-CS-25 as per EASA.IM.A.191 for DHC-8-400 Series. The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection (...) | Installation must be in accordance with SaTGlo MDL, Report Number M07-2, Rev. H, dated 1 August 2007. Operation must be in accordance with AFM Supplement SaTGlo Report No.M07-11, Rev. B, dated 25 October 2004, and Flight Attendant Manual Supplement, S (...) | Cabin interior arrangement must be approved in accordance with SaTGlo MDL, Report Number M07-2, Rev. H, dated 1 August 2007 incorporating SaTGlo drawing EG-DHC8, Rev. E, dated 1 August 2007. Electrically-lit door exit marker lighting levels and cabin (...) | 16.03.2011 |
| 10040330 | STG AEROSPACE Inc. | | REV. 1 | FAA STC ST03684AT | 777-200/300/300ER | Validation of FAA STC ST03684AT (Installation of SaTGlo Photoluminescent Escape Path Marking System) | The Certification Basis for the original product remains applicable to this certificate/ approval. | FAA STC ST03684AT, Amendment November 3, 2011 STG report AC37-2 Rev. F (Master Drawing List) AFM supplement AC37-11 Rev. B (Airplane Flight Manual Supplement) Or later revisions of the above listed documents approved by EASA in accordance with EASA ED De (...) | none | 25.07.2012 |
| 10042956 | STG AEROSPACE Inc. | EASA.IM.A.S.01162 | REV. 1 | FAA STC ST02292AT | A318-111 / -112 / -121 / -122 A319-111 / -112 / -113 / -114 A319-115 / -131 / -132 / -133 A320-111 / -211 / -212 A320-214 / -215 / -216 A320-231 / -232 / -233 A321-111 / -112 / -131 / -211 A321-212 / -213 / -231 / -232 | Installation of SaTGlo photoluminescent floor proximity emergency escape path marking system (PFEPPMS) - Validation of FAA STC N ST02292AT | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA STC ST02292AT, amended May 25, 2011 Certification Plan 0713186000, Rev. A, dated March 16, 2011 Master Drawing List L67-2, Rev. L, dated October 18, 2011 Instructions for Continued Airworthiness L67-10, Rev. G, dated November 18, 2010 EWEIS ICA 071318 (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 06.06.2013 |
| 10039197 | STRAIGHT FLIGHT, Inc. | | | | CESSNA 510 | Modification of the aircraft interior to allow for two configurations: 1) Observer configuration and 2) back to the Type Certified standard Executive configuration in accordance with Straight Flight Master Drawing List Document No.4979-MDL-01, Revise (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List: Doc. No. 4979-MDL-01, Revision B, dated June 15, 2010. Aircraft Flight Manual Supplement: Doc No. FMS-4979-R007, Rev. N/C, dated June 16, 2010 Instructions for Continued Airworthiness: Doc No. 4979-R-006, dated June 7, 2010, Revision A (...) | According limitations section of AFMS. This approval is limited to the installation and use of a 40 Cubic Foot occupant Oxygen supply tankonly. | 17.04.2012 |
| 10015479 | STRUCTURAL INTEGRITY ENGINEERING | EASA.IM.A.S.01233 | | SA6073NM | BEECH 99 BEECH 99A BEECH 99B BEECH C99 | Structural Life Extension of Wing Spars according to FAA STC SA6073NM as amended on 30 December 1993. | | FAA Approved Airplane Flight Manual Supplements: Beechcraft C99 with outboard wing main spar assembly (P/N 99-110045) and Centre Section Lower Main Spar Cap (P/N 109-120016-5), Report SIE-12-601 Beechcraft 99, 99A and B99 with Kit Nos. 99-4023-1S or (...) | According to FAA STC SA6073NM and as further defined below. Compliance with the Airworthiness Limitations contained in the above referenced documents extends the existing 20,000 flight hour Beech fatigue life limits for wing spars of Beechcraft 99, 9 (...) | 02.01.2006 |
| 10045275 | SUPERJET INTERNATIONAL S.P.A. | | REV. 1 | | RRJ-95B | CN-IJ93-0022 - LAVATORIES - ADD REINFORCEMENT ON LAVATORIES STRUCTURE AND UPDATE AFM | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | as defined in certification plan OPS-CED-13143 Rev B dated 02/07/2013 Change Notice CN-IJ93-0022 Rev A dated 27/06/2013 Change Approval Sheet OPS-CDC-13166 Rev A dated 24/07/2013 Equipment Qualification List OPS-CDC-13043 Rev C dated 22/07/2013 AFM Sup (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 25.07.2013 |
| 10033267 | SUPERVAN SYSTEMS, LTD. | | REV. 2 | FAA STC SA10841SC | 208 AND 208B | Installation of the Honeywell TPE331-12JR engine and Hartzell HC-B4TN-SQL/LT10890-NK (de-ice) or HC-B4TN-SNL/LT10890-N (non de-ice) or Hartzell propeller HC-B4TN-SQL/LT10891NK (de-ice) or HC-B4TN-SNL/LT10891N (non de-ice) propeller and associated comp (...) | The certification basis for the Cessna 208 and 208B modified in accordance with Supervan Master Data List, SSL-MDL-208-4001, Rev. D, dated May 22, 2012, or later FAA approved revision is as follows: a) The type certification basis for the Cessna 208B (...) | A) Master Drawing List SSL-MDL-208-4001, Rev. C, dated June 10, 2009B Instructions for Continued Airworthiness Document 9004-21, Rev. IR, dated September 15, 2008, including Airworthiness Limitation Section, FAA approved November 26, 2008C Flight Manu (...) | This modification is not approved on airplanes with the TKSLce Protection system. For Cessna 208: This STC is limited to standard version (without cargo pod) to serial numbers 20800001 through 20800415 and 20800417 through 20800499 For Cessna 208B: This S (...) | 19.07.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|----------------------|--|--|--|---|---|---------------|
| 10014781 | SWS CERTIFICATION SERVICES LTD | EASA.A.S.02979 | | | AIRBUS A320 SERIES BOEING 717 BOEING 727 BOEING 737 BOEING 747 BOEING 757 BOEING 767 BOEING DC-9 EMBRAER ERJ 135 SERIES EMBRAER ERJ 145 SERIES MD-11 | Installation of STG Aerospace SaTGlo Photoluminescent Bulkhead/ Divider Exit Signs. | | As listed in SWS Certification Ltd. Design report 3001-DR-001, Issue 1 or later EASA approved issue. | As defined in SWS Certification Ltd. Design report 3001-DR-001, Issue 1 or later EASA approved issue. | 19.06.2007 |
| 10015139 | SWS CERTIFICATION SERVICES LTD | EASA.A.S.03495 | | | BAE 146 | STG Aerospace Wireless Emergency Primary Power System (WEPPS) Installation. | | - Certification Plan 3011-CEP-01.- Compliance Report 3011-COM-01 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 08.12.2008 |
| 10032329 | SWS CERTIFICATION SERVICES LTD | | REV. 1 | | BOEING B747-300 BOEING B747-400 | Installation of SaTGlo Inc. Photoluminescent Floor Proximity Emergency/Escapes Path Marking System (FPPEPMS) in accordance with SWS Certification Services Design Report 3046-DR-01 Rev. 2, dated 27 August 2010 or later approved revision. | The Certification Basis for the original product remains applicable to this certificate/ approval. | 3046-MMS-01 Issue 1, Maintenance Manual Supplement 3046-CMS-01 Issue 1, Crew Manual Supplement 3046-SB-01 Issue 1, Service Bulletin or later revisions of the above listed documents approved by EASA | none | 08.11.2010 |
| 10040318 | SWS CERTIFICATION SERVICES LTD | | REV. 2 | | A340-312 | This modification covers:- Installation of Recaro 3410 785 Economy (E/C) seats and Avio Business Class (B/C) seats, fitted with Panasonic EX2 IFE, onto TAP A340-300 aircraft.- Installation of AmSafe Airbag System to demonstrate compliance to CS25.562(c (...)) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Compliance Report 3091-COM-01 Issue 1 <-,-> dated 12 Oct 2012.- Maintenance Manual 3091-MMS-01 Issue 1, dated 12 Oct 2012.- Declaration of compliance SWS-C-3091 Issue A, dated 16 Oct 2012 or later revisions of the above listed documents approved by EASA | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 30.04.2013 |
| 10040913 | SWS CERTIFICATION SERVICES LTD | | REV. 1 | | B747-8 | Installation of the Aeroloft(TM) Overhead Sleeping Compartment immediately forward of the Overhead Crew Rest. Two configurations are certified: 1. With lounge module 2. Without lounge module and with following peculiarities: 2.1 Stair module isolated aft (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.- CRI A-100 EASA Certification basis definition- CRI H-01 Special condition for Enhance (...) | Certification Plan 3069-CEP-01 Issue 02 Certification Plan 3090-CEP-01 Issue 01 Master Configuration Scheme 3069-MCS-01 Issue 01 Master Configuration Scheme 3090-MCS-01 Issue 01 | No passenger allowed in the cabin. | 11.12.2012 |
| 10015622 | TACA INTERNATIONAL AIRLINES | EASA.IM.A.S.01574 | | | AIRBUS A320-233 | Automatic ELT Provisions installation and activation | | Grupo TACA Engineering Order A320MO25-089/99 Revision dated July 31, 2000 | This STC is limited to A320 MSN 561 | 20.10.2005 |
| 10015623 | TACA INTERNATIONAL AIRLINES | EASA.IM.A.S.01575 | | | AIRBUS A320-233 | Automatic ELT- Provisions installation and activation | | Grupo TACA Engineering Order A320MO25-089/99, Revision 3, dated June 7, 2001 | This STC is limited to A320-233 MSN 558 | 20.10.2005 |
| 10015624 | TACA INTERNATIONAL AIRLINES | EASA.IM.A.S.01576 | | | A320-233 | Equipment and furnishing - delivery configuration of passenger cabin - 180 pax | | Taca Engineering Order A320IN25-122/05, Revision 0, dated August 31, 2005. | This STC is limited to the A320 MSN 561 | 26.05.2005 |
| 10015667 | TACA INTERNATIONAL AIRLINES | EASA.IM.A.S.01643 | | | AIRBUS A320-233 | Equipment and furnishing - Delivery configuration of passenger cabin - 180 pax | | Taca Engineering Order A320IN25-121/05, Revision 0, dated August 31, 2005. | This STC is Limited to the A320 MSN 558. | 26.10.2005 |
| 10015703 | TACA INTERNATIONAL AIRLINES | EASA.IM.A.S.01690 | | | AIRBUS A320-233 | Autoflight - Activate FMGC Nav data base to 400k words - Pin programming | | TACA Engineering Order A320MO22-007/02, revision 1, dated August 29, 2002 | 1. This STC is limited to the A320 MSN's 558 and 561 | 04.11.2005 |
| 10032788 | TALON AEROSPACE LLC | | | FAA STC ST03574AT | SEE EASA APPROVED MODEL LIST | Anti Collision Light installation This STC installs a new model of Anti Collision Light. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CS-25 Amendment 4 Special Condition H-01 Issue 1 (ICA on EWIS) The requirements for envi (...) | Master Data List, ref: MDL-001, Revision 7- Airbus A300 / A310: Installation Instructions A0-001 Revision 2, dated 01 October 2010- Airbus A318 / A319 / A320 / A321: Installation Instructions A2-001 Revision 2, dated 01 October 2010- Airbus A330 / A340: (...) | None | 01.12.2010 |
| 10042345 | TAMARACK AEROSPACE GROUP, INC. | | | FAA STC SA02216SE | SR20/SR22 | Removal of mechanical link between engine throttle control and propeller RPM control. Installation of a separate RPM control lever, linkage between Cirrus RPM cable/cam followers, center console to accommodate three-lever configuration, new knobs for (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List: TAG-CSQ-301, revision B, dated March 7, 2012; Installation Manual/ Instructions for Continued Airworthiness Supplement: TAG-CSQ-203, revision IR, dated January 30, 2012; Airplane Flight Manual Supplement: TAG-CSQ-201, revision IR, (...) | SR20 serial numbers 1005 through 1885; SR22 serial numbers 0002 and subsequent without Garmin Perspective Avionics Installation; SR20 airplanes equipped with Hartzell BHC-J2YF-1BF/F7694 propeller assemblies must have a placard installed reading "Avoid o (...) | 26.11.2012 |
| 10043927 | TAMARACK AEROSPACE GROUP, INC. | | | FAA STC SA02247SE | SR20, SR22, SR22T | Cargo Conversion Removal of rear seats and installation of cargo shelf and restraints system in accordance with FAA STC SA02247SE. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Removal of rear seats and installation of cargo shelf and restraints system controlled by Cirrus Master Drawing List (MDL) TAG-CRT-205, Revision B, dated October 23, 2012; Installation in accordance with Cirrus Installation Instructions/Instructions for (...) | This STC is applicable to the following serial blocks of Cirrus Design aircraft:- Cirrus SR20: Serial Numbers 1005 and subsequent.- Cirrus SR22: Serial Numbers 0002 and subsequent.- Cirrus SR22T: Serial Numbers 0001 and subsequent. Prior to installation (...) | 05.03.2013 |
| 10030951 | TASS Inc. | | | FAA STC ST00593SE-D | B747-400 | Active Noise Reduction Headset Installation Provisions in accordance with BF Goodrich Master Drawing List D2000-396, sheet 1, Revision New, dated November 22, 2000 or later FAA/DAS approved revision. Validation of FAA STC ST00593SE-D | The Certification Basis for the original product remains applicable to this certificate/ approval. | None | None | 15.07.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------|------------------|----------------|----------------------|---|---|--|---|---|---------------|
| 10031609 | TASS Inc. | | | FAA STC ST00583SE-D | B757-200 | Installation of Passenger Accommodations - STC ST00583SE-D | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA Statement of Compliance reference 1505-10-233 dated 18 May 2010. FAA STC No. ST00583SE-D dated 07 May 2010. BF Goodrich Master Data List 99-060, pages 1-20, Revision A dated 30 July 1999. TASS Inc document TASS-BAB-10-GQ001, Revision 00 dated 17 (...) | None | 30.08.2010 |
| 10044903 | TASS Inc. | | | FAA STC ST00970SE-D | 737-700 | Installation of avionics equipment according to Virgin (VOZ) fleet standards. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | GOODRICH Master Data List (MDL) 2002-167 Rev C Dated August 19th, 2002 GOODRICH Supplemental Instructions for Continued Airworthiness (ICA) D2002-205 dated July 01st, 2002 TASS EU Certification Plan C737-00-0340-CP-01 dated April 23rd, 2013 or later rev (...) | A copy of this certificate and GOODRICH MDL 2002-167 rev C dated August 19th, 2002 (or later approved version) must be maintained as part as the permanent records for the modified aircraft. The supplemental ICA GOODRICH D2002-205 dated July 01st, 2002 (...) | 16.05.2013 |
| 10014381 | TATENHILL AVIATION Ltd. | EASA.A.S.01665 | | | PILATUS PC-12/45 | Substantiation of Garmin AT GPS400 installation for IFR | | Tatenhill Aviation Ltd Modification TAL_TAD 019/05 Tatenhill Aviation AFM Supplement TAL-BRNAV FMS-TAD 019/05 Issue 1 or later approved revision. | | 20.06.2006 |
| 10014412 | TATENHILL AVIATION Ltd. | EASA.A.S.01711 | | | CESSNA F182-Q | Design data and equipment installation in order to approve the aircraft for IFR. | | Flight Manual Supplement TAL-BRNAV FMS-TAD 002/06 Issue 1 | -- | 08.01.2007 |
| 10014909 | TATENHILL AVIATION Ltd. | EASA.A.S.03153 | | | CESSNA 500 | Approval of a Mode S EHS ATC System. | | Tatenhill Aviation Limited Change TAL-TAD 020/07 Flight Manual Supplement FMS.02007 Revision A or later EASA approved revision | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 10.07.2008 |
| 10014923 | TATENHILL AVIATION Ltd. | EASA.A.S.03170 | | | CESSNA 500 | Approval of a Mode S EHS ATC System. | | Tatenhill Aviation Limited Change TAL-TAD 019/07 Flight Manual Supplement FMS.01907 Revision A or later EASA approved revision | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 23.09.2008 |
| 10015168 | TATENHILL AVIATION Ltd. | EASA.A.S.03538 | | | CESSNA 525 | Dual Garmin GPSNAV/COM Upgrade. | | Mod. Ref. TAL-TAD 005/08 MDL.00508, Rev. AAFM Supplement - FMS.00508 Rev.A | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 26.05.2008 |
| 10017132 | TATENHILL AVIATION Ltd. | EASA.R.S.01225 | | | AGUSTA A-109 C | Design Data for Compliance with UK CAA Generic requirement 4 by Installation of a battery Low Voltage Indicator | | Tatenhill Aviation Ltd Modification TAL-TAD 025/05 Tatenhill Aviation Ltd RFM supplement TAL-TAD FMS 3 | | 07.04.2006 |
| 10017133 | TATENHILL AVIATION Ltd. | EASA.R.S.01227 | | | AGUSTA A-109 All | Approval of Low Voltage Warning Light and Associated Flight Manual Supplement | | Flight Manual Supplement: TAL - TAD FMS-2 Major Change Design Forms: TAL - TAD 024/05, issue 1 or later approved revision | Agusta A109 All Series | 31.08.2006 |
| 10017208 | TATENHILL AVIATION Ltd. | EASA.R.S.01360 | | | AGUSTA A-109 All | Approval of replacement #TOT# gauges | | 2. Master Document List MDL.01007, Rev A, dated 02.03.2009, Flight Manual Supplement FMS 01007, Revision A, dated 07.08.2008 | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 06.03.2009 |
| 10017220 | TATENHILL AVIATION Ltd. | EASA.R.S.01377 | | | AGUSTA A-109 | Design data compliance with Generic Requirement No. 4. | | Tatenhill Aviation Master Document List TAL-TAD 015/07 Tatenhill Aviation AFM Supplement TAL-TAD FMS 18 Revision A. | As per AFM Supplement TAL-TAD FMS 18 Revision A. | 26.07.2007 |
| 10026831 | TATENHILL AVIATION Ltd. | P-EASA.A.S.3878 | | | PA-24-400 PA-24, PA-24-250, PA-24-260, | Approval of an Aspen EFD1000 Primary Flight Display (PFD) | CS 23 | -Master Document List: MDL.00809-Airplane Flight Manual Supplement FMS.00809 Rev A or later EASA approved version-Instructions for Continued Airworthiness ICA.00809 | STC approved for Piper PA24 series | 13.08.2009 |
| 10026846 | TATENHILL AVIATION Ltd. | P-EASA.A.S.03879 | | | F182-P, F182-Q, FR-182 | Approval of an Aspen EFD1000 Primary Flight Display (PFD) | CS 23 | Master Document List: MDL.00609-Airplane Flight Manual Supplement FMS.00609 Rev A or later EASA approved version-Instructions for Continued Airworthiness ICA.00609 | STC approved for F182P, F182Q and FR182 series types only | 17.08.2009 |
| 10028330 | TATENHILL AVIATION Ltd. | P-EASA.A.S.03956 | | | BEECH 200 | Design data to provide European approval of FAA STC: SA5254NM | IM.A.277 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval FAA AC.23.1309-1CFAA AC.23.1311-1BINT/POL/23/1INT/POL/23/3The certificated noise (...) | Master Data List TAL-TAD 014/09 Rev B. Instructions for Continued Airworthiness ICA 01409 Rev B. Flight Manual Supplement 01409 Rev A or later revisions of the above listed documents approved by EASA | Serial number BB-877 only | 17.12.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|-------------------|----------------|----------------------|---|---|---|---|---|---------------|
| 10029450 | TATENHILL AVIATION Ltd. | | | | M20, M20A, M20B, M20C, M20D M20E, M20F, M20G, M20J, M20K M20L, M20R, M20S, M20TN | Installation of Aspen EFD1000 Primary Flight Display (PFD) | The Certification Basis of the original product and the following additional / alternative airworthiness requirements are applicable to this change: CS-23 has been elected to comply with for those areas affected by the change unless otherwise stated EASA (...) | EASA approved Design Report with Installation Instructions, Doc. No. TAD.02209, Revision B, dated 23-Mar-2010, or later approved revision EASA approved Airplane Flight Manual Supplement, Doc. No. FMS.02209, Revision A, dated 20-Nov-2009, or later approve (...) | This change is limited to the configuration of aircraft, equipment type, functions and software versions as listed in EASA approved Design Report, Doc. No. TAD.02209, Revision B, dated 23-Mar-2010 | 25.03.2010 |
| 10033913 | TATENHILL AVIATION Ltd. | | | | SEE EASA APPROVED MODEL LIST | Design Data for the Approval of a Honeywell KFD840 Primary Flight Display (PFD). | CS-23 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Manual IM.00910 Revision C Master Document List MDL.00910 Revision A Instructions for Continued Airworthiness ICA.00910 Revision A Flight Manual Supplement FMS.00910 Revision A or later revisions of the above listed documents approved by EASA. | None | 17.02.2011 |
| 10034482 | TATENHILL AVIATION Ltd. | | | | SEE EASA APPROVED MODEL LIST | Design Data for the Approval of a Honeywell KFD840 Primary Flight Display (PFD). | CS-23 The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Manual IM.01510 Revision A Master Document List MDL.01510 Revision B Instructions for Continued Airworthiness ICA.01510 Revision A Flight Manual Supplement FMS.01510 Revision A or later revisions of the above listed documents approved by EASA. | None | 07.04.2011 |
| 10035157 | TATENHILL AVIATION Ltd. | | | | 1. PA-44-180, PA-44-180T 2. PA-34-220T 3. PA-34-200, PA-34-200T | Design data for the approval of a Honeywell KFD840 Primary Flight Display (PFD) | CS-23 The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Manual IM.02510 Revision A Master Document List MDL.02510 Revision A Instructions for Continued Airworthiness ICA.02510 Revision A Flight Manual Supplement FMS.02510 Revision A or later revisions of the above listed documents approved by EASA. | None | 30.05.2011 |
| 10036731 | TATENHILL AVIATION Ltd. | | | | SEE EASA APPROVED MODEL LIST | Design Data for the approval of a Honeywell KFD840 Primary Flight Display (PFD). | CS-23 Amendment 1 The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Installation Manual IM.02410 Revision A, dated 09 July 2011. Master Document List MDL.02410 Revision B, dated 27 September 2011. Instructions for Continued Airworthiness ICA.02410 Revision A, dated 09 July 2011. Flight Manual Supplement FMS.02410 Revisi (...) | None | 04.10.2011 |
| 10039139 | TATENHILL AVIATION Ltd. | | | | 1. PA-34-220T 2. PA-34-200, PA-34-200T 3. PA-34-220T 4. PA-44-180, PA-44-180T | Approval of a Transponder System in accordance with AMC 20-2 | CS23 Amendment 2 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate approval. The requirements for environmental protection and the associated certified (...) | Installation Manual IM.00611 Revision A, dated 16 January 2012. Master Document List MDL.00611, Revision A, dated 29 March 2012. Instructions for Continued Airworthiness ICA.00611, Revision A, dated 02 March 2012. Flight Manual Supplement FMS.00611, Rev (...) | None | 12.04.2012 |
| 10045284 | TATENHILL AVIATION Ltd. | | | | PA-31 | Approval of an Aspen EFD-1000 Primary Flight Display against the requirements of EU-OPS for Dual Pilot Operations. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 23 Amendment 3. CS 23.611 CS 23.771 CS 23.777 CS 23.867 CS 23.1301 CS 23.1303 CS 23.1309 CS 23.1311 CS 23.1321 CS 23.1322 CS 23.1 (...) | Master Document List MDL.00812 Revision B Installation Manual IM.00812 Revision A Instructions for Continued Airworthiness ICA.00812 Revision A Flight Manual Supplement FMS.00812 Revision A or later revisions of the above listed documents approved by EASA. | None | 12.06.2013 |
| 10041362 | TC INTER-INFORMATICS A.S. | | REV. 1 | | 737-800 | D27 Cabin layout reconfiguration Revision 1: major update of the Instructions for Continued Airworthiness document D27-D01. Associated technical documentation and STC limitations updated. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Definition document, reference D27-D01 revision 5, dated 30 October 2012. Project Compliance Checklist, reference D27-D03 revision 2, dated 30 October 2012. Declaration of Compliance, reference D27-D05 revision 5, dated 30 October 2012. FCOM Supplement to D6- (...) | Limitations/ Applicability of this STC is limited to the aircraft serial numbers listed in Definition Document D27-D01. Conditions: Prior to installation of this STC it must be determined that the relationship between this STC and any other previously (...) | 05.02.2013 |
| 10015625 | TDG AEROSPACE INC | EASA.IM.A.S.01577 | | FAA STC SA6042NM | DC-9-81 (MD-81) DC-9-82 (CF6) DC-9-83 (MD-83) DC-9-87 (MD-87) MD-11 | Installation of NOFOD Overwing Anti-Ice System | | TDG Aerospace Drawing List No E93-104, Rev U dated 31 October 2002 | 1. Flight Manual Supplement TDG Doc No E93-096 Rev D dated 27 September or later approved revision 2. Instructions for Continued Airworthiness TDG Doc E93-109 Revision P dated 11 February 2003. This STC is approved only for the product configuration (...) | 05.09.2005 |
| 10016121 | TDG AEROSPACE INC | EASA.IM.A.S.02314 | REV. 1 | FAA STC ST01950LA | BOEING 757-200 BOEING 757-300 | Universal Fault Interrupter (UFI) Installation | | TDG Aerospace, MDL 10-3200-0009 (rev C, 11th January 2007) AFM Supplement 20-3200-003 (10th January 2007) | The UFI is a Critical Design Control Configuration Limitation (CDCL) item. Its operation and instructions for maintenance and inspection must be performed in accordance with the TDG Instructions for Continued Airworthiness (document 20-3200-0001, r (...) | 19.04.2007 |
| 10016450 | TDG AEROSPACE INC | EASA.IM.A.S.02796 | | FAA STC ST02076LA | BOEING 737-600 BOEING 737-700 BOEING 737-700C BOEING 737-800 BOEING 737-900 | Universal Fault Interrupter Installation Kit for B737NG. | | FAA approved Airplane Flight Manual Supplement (Ref. TDG document 20-3202-0003), Aircraft Maintenance Manual (Ref. TDG document 20-3202-0001), Aircraft Wiring Diagrams Manual (Ref. TDG document 20-3202-0005), Aircraft Illustrated Parts Catalogue (Re (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 25.08.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------|-------------------|----------------|----------------------|---|--|--|---|--|---------------|
| 10032604 | TDG AEROSPACE INC | | | FAA STC ST02285LA | B767-200, -300, -300F | Installation of Universal Fault Interrupter (UFI) for Centre Wing Fuel Tank Electrical Fault Protection. | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this approval. | UFI 767 Installation Kit Master Drawing List, 10-3206-0002, Revision B, 18/03/2010 Airplane Flight Manual Supplement, 20-3206-0002, 28/02/2010 Instructions for Continued Airworthiness, 20-3206-0001, Revision E, 04/02/2010 (EWIS Compliant), or later revision (...) | If the aeroplane has supplemental (high flow) jetison pumps installed, the installation of this STC requires them to be deactivated. This STC approves accomplishment of the deactivation procedures for the supplemental (high flow) jetison pumps (design (...)) | 17.11.2010 |
| 10016729 | TECH-TOOL PLASTICS, Inc. | EASA.IM.R.S.01032 | | FAA STC NO SR03016AT | BELL 407 | Installation of windows for Bell 407 Model Helicopters in accordance with:-the FAA STC SR03016AT dated 7/1/2005.-Master Drawing List 407-1000-100, dated 23/2/2004. | | Window installation 407-2000 dated 11/2/2004 Window Assy, Windshield 407-1001 dated 10/10/2003 Window Assy, Lower Nose 407-1002 dated 10/7/2003 Window Assy, Lower Nose Quick Change 407QC-1002 dated 9/10/2003 Window Assy, Skylight 407-1003 dated 10/10/1 (...) | 1. The compatibility of this design change with other previously approved modifications must be determined by the installer. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the par (...) | 20.09.2005 |
| 10030363 | TECH-TOOL PLASTICS, Inc. | | | FAA STC SR09127RC-D | AS350 B, AS350 B1, AS350 B2 AS350 B3, AS350 BA, AS350 D | Installation of a Vertical Sliding Window in the Right and/or Left Hand Sliding Door FAA STC SR09127RC-D | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Drawing List 350A82-1750, Rev B, dated August 25, 2000- Rotorcraft Flight Manual Supplement (FMS) 350FMS-066, Revision B, dated July 24, 2008- ICA, Doc Number TTPICA-350-001, dated July 27, 1999 or later revisions of the above listed documents approved (...) | None | 11.06.2010 |
| 10036624 | TECH-TOOL PLASTICS, Inc. | | | FAA STC SR03641AT | EC 120 B | Replace the existing Eurocopter EC120 B windows with Tech-Tool Plastics, Inc. replacement windows. (Note: This approval constitutes the EASA validation of the FAA STCSR03641AT, issued by the FAA on June 30, 2011) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Drawing List (MDL) EC120/130-2000-100, Revision C dated October 31, 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) Eurocopter EC 12 (...) | None | 23.09.2011 |
| 10037086 | TECH-TOOL PLASTICS, Inc. | | | FAA STC SR03640AT | EC130 B4 | Installation of Replacement Windows This STC is the validation of FAA STC SR03640AT | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Tech-Tool Plastics, Inc. Master Drawing List (MDL), EC120/130-2000-100, Revision C dated October 31, 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None | 31.10.2011 |
| 10037490 | TECH-TOOL PLASTICS, Inc. | | | FAA STC SR09133RC-D | AS 355 E AS 355 F AS 355 F1 AS 355 F2 AS 355 N | Installation of a Vertical Sliding Window in the Right Hand or Left Hand Sliding Door in accordance with Drawing List 350A82-1750, Revision B, dated 08/25/00. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | RFM 355FMS-020, dated March 19, 2001 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). Drawing List 350A82-1750, Revision B, dated 08/25/00. | Both configurations cannot be installed concurrently (Right Hand and Left Hand). | 30.11.2011 |
| 10038180 | TECH-TOOL PLASTICS, Inc. | | | FAA STC SH5822SW | AS355E, AS355F, AS355F1 AS355F2, AS355N | Installation of Tech-Tool performed acrylic plastic windshield in accordance with Tech-Tool Drawing List (Doc 3501-2023) No 1, Rev A, dated 08/28/09 (FAA STC SH5822SW) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Tech-Tool Drawing List (Doc 3501-2023) No 1, Rev A, dated 08/28/09 | None | 26.01.2012 |
| 10038186 | TECH-TOOL PLASTICS, Inc. | | | FAA STC SH5797SW | AS350B, AS350B1, AS350B2 AS350B3, AS350BA, AS350D | Installation of Tech-Tool performed acrylic plastic windshield in accordance with Tech-Tool Drawing List (Doc 3501-2023) No 1, Rev A, dated 08/28/09 (FAA STC SH5797SW) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Tech-Tool Drawing List (Doc 3501-2023) No 1, Rev A, dated 08/28/09 | None | 26.01.2012 |
| 10038761 | TECH-TOOL PLASTICS, Inc. | | | FAA STC SR09046RC | AS 350 B, BA, B1, B2, B3, D | Installation of a reverse slide camera window | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Drawing List No. 3501-2035-100<-(-<-> Rev. A, dated 01/31/12 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 12.03.2012 |
| 10038762 | TECH-TOOL PLASTICS, Inc. | | | FAA STC SR09045RC | AS 355 E, F, F1, F2, N | Installation of a reverse slide camera window | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Master Drawing List No. 3501-2035-100<-(-<-> Rev. A, dated 01/31/12 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 12.03.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------|---------------------|----------------|----------------------|--|---|--|--|--|---------------|
| 10039763 | TECH-TOOL PLASTICS, Inc. | EASA.IM.R.S.0153 | REV. 1 | | MBB-BK 117 A-1 MBB-BK 117 A-3 MBB-BK 117 A-4 MBB-BK 117 B-1 MBB-BK 117 B-2 MBB-BK 117 C-1 MBB-BK 117 C-2 | Installation of Tech-Tool Plastics, Inc. Replacement Windows in EUROCOPTER MBB-BK 117 Installation of windows in accordance with Tech-Tool Plastics, Inc. Master Drawing List No. BK117-1000-100 i.a.w. FAA STC No. SR0267AT | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Tech-Tool Plastics, Inc. Master Drawing List No. BK117-1000-100, dated 23 September 2003, or later approved revisions. | None. | 22.05.2012 |
| 10039788 | TECH-TOOL PLASTICS, Inc. | | REV. 1 | FAA STC SH8755SW | AS 355 E AS 355 F AS 355 F1 AS 355 F2 AS 355 N | Installation of aft Cabin Comfort Windows | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List No. 3501CC-2027-100, Rev A, dated 02/04/12 | none | 25.05.2012 |
| 10039789 | TECH-TOOL PLASTICS, Inc. | | | FAA STC SH8754SW | AS 350 B1 AS 350 B2 AS 350 B3 AS 350 BA AS 350 D AS350 B | Installation of aft Cabin Comfort Windows | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List No. 3501CC-2027-100, Rev A dated 02/04/12 | none | 23.05.2012 |
| 10042997 | TECH-TOOL PLASTICS, Inc. | LBA RC 1108 | | FAA STC SR09000RC | AS350B/BA/B1/B2/B3 | Installation of miscellaneous windows in accordance with Tech-Tool Plastics, Inc. MDL 3501-2000-1. This STC is the re-issuance of LBA-Germany STC RC 1108 issued on 27 September 1999. | | MDL No. 3501-2000-1, Rev. A, dated 21/08/1998 | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 08.07.2013 |
| 10042998 | TECH-TOOL PLASTICS, Inc. | LBA RC 1117 | | FAA STC SR09003RC | AS 355 E/F1/F2/N | Installation of miscellaneous windows in accordance with Tech-Tool Plastics, Inc. MDL 3501-2000-1. This STC is the re-issuance of LBA-Germany STC RC 1117 issued on 27 September 1999. | | MDL No. 3501-2000-1, Rev. A, dated 21/08/1998 | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 09.07.2013 |
| 10042999 | TECH-TOOL PLASTICS, Inc. | LBA RC 1099 | | FAA STC SH8967SW | AS355E/F1/F2/N | Installation of aft Cabin Comfort Windows in accordance with Tech-Tool Plastics, Inc. MDL 3501CC-2032-100. This STC is the re-issuance of LBA-Germany STC RC 1113 issued on 27 September 1999. | | MDL No. 3501CC-2032-100, Rev. I.R., dated 15/07/1994 | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 09.07.2013 |
| 10043000 | TECH-TOOL PLASTICS, Inc. | LBA RC 1113 | | FAA STC SH8966SW | AS350B/BA/B1/B2/B3 | Installation of aft Cabin Comfort Windows in accordance with Tech-Tool Plastics, Inc. MDL 3501CC-2032-100. This STC is the re-issuance of LBA-Germany STC RC 1099 issued on 27 September 1999. | | -MDL No. 3501CC-2032-100, Rev. I/R dated 15/07/1994-Tech-Tool Plastics, Inc. Installation Instructions for Part Number 3501CC-2032-3, -4, -6 Aft Sliding Cabin Comfort Windows, Report TTP-IRev. I/R - 07/11/94 | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 09.07.2013 |
| 10014414 | TELAIR INTERNATIONAL AB | EASA.A.S.01713 | | | AIRBUS A319 AIRBUS A320 AIRBUS A321 | Installation of Sliding Carpet Loading System in Airbus A319/320/321 in accordance with Telair International AB Master Document List (MDL) | | 1) Telair International AB document "Master Document List (MDL) No T-D090, revision A, dated December 14, 2005 or later revision (A321)2) Telair International AB document "Master Document List (MDL) No T-D065, revision A, dated April 4, 2004 or later revision (A319/320) | N/A | 13.02.2006 |
| 10016023 | TELAIR INTERNATIONAL Inc. | EASA.IM.A.S.02191 | REV. 1 | FAA STC ST00837 LA | BOEING 747-400 BOEING 747-400F | Installation of a Cargo Handling System on the lower deck of the B747-400, -400F | | FAA approved Telair International Index List IL 192100 Rev.C dated May 24, 1999 or later FAA approved revisions. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The i (...) | 12.01.2009 |
| 10016031 | TELAIR INTERNATIONAL Inc. | EASA.IM.A.S.02200 | REV. 1 | FAA STC ST00459LA | BOEING 747-400SF | Installation of a Cargo Handling System on the main deck of the B747-400SF in accordance with FAA approved Telair International Index List ER 193100 Rev. N/C dated October 21, 1997 or later approved revisions. Validation of FAA STC ST00459LA | | Weight & Balance Control And Loading Manual Report Number 25-55-67 Revision #for Main Deck Cargo Handling System System Number 193100 | 1. The installation of this modification by third persons is subject to written permission of the approval holder. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other (...) | 11.12.2008 |
| 10036235 | TELAIR INTERNATIONAL Inc. | P-EASA.IM.A.S.03066 | REV. 1 | FAA STC 02365LA | B747-8F | Lower Deck Cargo Handling System, Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. EASA CRI A-100 The requirements for environmental protection and the associated certific (...) | PSCP 192100CPP-8F and FAA STC 02365LA or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 26.08.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|-------------------------|----------------|----------------------|---|--|---|---|---|---------------|
| 10036236 | TELAIR INTERNATIONAL Inc. | P- EASA.IM.A.S.03067 | REV. 1 | FAA STC 02367LA | B747-8F | Main Deck Cargo Handling System, Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval EASA CRI A-100 The requirements for environmental protection and the associated certificate (...) | PSCP 193100CPP-8F and FAA STC 02367LA or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 26.08.2011 |
| 10036941 | TELAIR INTERNATIONAL Inc. | | REV. 1 | FAA STC ST02367LA | 747-8F | Main Deck Cargo Handling System, Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval EASA CRI A-100 The requirements for environmental protection and the associated certificate (...) | PSCP 193100CPP-81F, FAA STC 14144LA-T and Index List 193100-8 Revision B or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 20.10.2011 |
| 10036942 | TELAIR INTERNATIONAL Inc. | | REV. 1 | FAA STC ST02365LA | 747-8F | Lower Deck Cargo Handling System, Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval EASA CRI A-100 The requirements for environmental protection and the associated certificate (...) | PSCP 192100CPP-705, FAA STC 14522LA-T and Index List 192100-8 Revision B or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 20.10.2011 |
| 10038077 | TELAIR INTERNATIONAL Inc. | | | FAA STC ST02365LA | 747-8 | Installation of a Lower Deck Cargo Handling System | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval EASA Certification Review Item "Telair Cargo Handling Systems/EASA Certification Basis D (...) | FAA STC Number ST02365LA dated 23.12.2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) FAA Letter Ref. 120L-11-613 dated 23.12.2011 | None | 17.01.2012 |
| 1004562 | TELAIR INTERNATIONAL Inc. | | | FAA STC ST02148LA | 767-300BCF | Installation of Cargo Handling System | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Telair International Index List, ER196100, Revision A, 05/06/2008 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement. | Only applicable to B767-300 PAX a/c modified to a Boeing Converted Freighter (B767-300BCF). Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design (...) | 04.07.2013 |
| 10015896 | TELEDYNE BATTERY PRODUCTS | EASA.IM.A.S.02016 | | FAA STC SA1013SO | BEECH B99 BEECH C90 BEECH E90 BEECH F90 | Replace existing nickel-cadmium battery with Gill BB638/T series lead-acid battery | | Installation instructions for replacing existing nickel-cadmium battery with Gill BB638/T series lead-acid battery, dated May 1, 1992, or later approved revisions. FAA Approved Airplane Flight Manual Supplement, dated August 3, 1978, (C90, E90, B99), (...) | 1) Beech C90 Serial Numbers LJ590 and after, E90 Serial Numbers LW58 and after, B99 Serial Numbers LA-2 and after, B99 Serial Numbers U146 and U148 through U164.2) FAA approved Airplane Flight Manual Supplement, dated August 3, 1978, (C90, E90, B99), (...) | 22.05.2006 |
| 10015897 | TELEDYNE BATTERY PRODUCTS | EASA.IM.A.S.02017 | | SA1016SO | BEECH 200 BEECH 200C BEECH 200CT BEECH 200T BEECH 300 BEECH 300LW BEECH B200 BEECH B200C BEECH B200T BEECH B300C | Replace existing nickel-cadmium battery with Teledyne G-6381E lead-acid battery or with Teledyne G-6381ES Sealed Recombinant lead-acid storage battery | | Installation instructions for replacing existing nickel-cadmium battery with Teledyne G-6381E lead-acid battery, dated August 3, 1978, or later approved revisions. Teledyne Master Data List 1016-7, Revision "NC", dated December 13, 2000, or later app (...) | FAA approved Teledyne Battery Products Airplane Flight Manual Supplement, dated August 3, 1978, for G-6381E battery, or March 21, 2001, for G-6381ES battery, or later approved revisions is required as part of this installation. | 23.05.2006 |
| 10015898 | TELEDYNE BATTERY PRODUCTS | EASA.IM.A.S.02018 | | SA1017SO | BEECH 65-90 BEECH 65-A90 BEECH 65-A90-1 BEECH A100 BEECH B90 BEECH C90 BEECH E90 | Replace existing nickel-cadmium battery with Teledyne Battery Model No. G-638E, G-6381E or G-6381ES | | Teledyne Battery Installation Instruction No. SA 1017SO, Revision A, dated April 7, 1998, or later approved revision. | | 23.05.2006 |
| 10016852 | TELEDYNE BATTERY PRODUCTS | EASA.IM.R.S.01228 | | SR01406LA | BELL 206A BELL 206B BELL 206L BELL 206L-1 BELL 206L-3 | Aircraft Battery Replacement with Alternative Designs # Replace existing nickel-cadmium batteries with Teledyne Battery Products Model G-641S battery | | Master Data List Document Revision No. NC dated March 12, 2002 or later approved revision Flight Manual Supplement Document No 206-3 Revision NC dated October 21, 2002 or later approved revision | | 14.09.2006 |
| 10016853 | TELEDYNE BATTERY PRODUCTS | EASA.IM.R.S.01229 | | SR01427LA | MD HELICOPTERS 369A MD HELICOPTERS 369D MD HELICOPTERS 369E MD HELICOPTERS 369F MD HELICOPTERS 369FF MD HELICOPTERS 369H MD HELICOPTERS 369HM MD HELICOPTERS 369HS MD HELICOPTERS 500N MD HELICOPTERS 600N | Aircraft Battery Replacement with Alternative Designs | | Master Data List Document No. 0500-02, revision A dated Dec 20, 2002 or later approved revision. FAA Approved Pilots Operating Handbook Supplement and Flight Manual Supplement No. MD500-03 FM dated February 27, 2003 or later approved revision | | 07.06.2006 |
| 10016928 | TELEDYNE BATTERY PRODUCTS | EASA.IM.R.S.01314 | | FAA STC ST1186SO | BELL 212 BELL 412 BELL 412 EP | Validation of FAA STC SH1186SO Aircraft Battery Replacement with Alternative Designs | | Master Part List TBPA-06-S-05 dated 21 October 1980 or Drawing Control Sheet 3708 dated 1 April 1987. Rotorcraft Flight Manual Supplement dated 25 September 1987 or later approved revisions. | | 20.06.2007 |
| 10027424 | TELEDYNE CONTROLS - FLIGHT DATA COM | P-EASA.A.S.03401 | | | MD-80 SERIES | Installation of Wireless Quick Access Recorder. | FAR 25. | Master Drawing List MOD-MDL-022-02 Issue 1 dated 30 June 2009. | None. | 12.09.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-------------------------------------|---------------------|----------------|----------------------|---|---|---|--|---|---------------|
| 10033194 | TELEDYNE CONTROLS - FLIGHT DATA COM | P-EASA.A.S.03937 | | | B737-600/-700/-800/-900/900ER | Installation of enhanced airborne data loader on Boeing 737seriesaircraft | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval-CS 25.1713 Amndt 5 - EWIS Fire Protection, Wire Insulation flammability-CRI H-1 Enhance (...) | Master Data List MOD-MDL-024-02 Issue 2 dtd. 12.Oct 2010-Instructions for Continued Airworthiness MOD-ICA-024 Rev. 1 dtd. 12.Feb2010or later revisions of the above listed documents approved by EASA | This modification is only applicable to 737 aircraft that already havean ADL installed or have provisions for installation of an ADL. Compatibility must be established for each aircraft. | 22.12.2010 |
| 10045436 | TELEDYNE CONTROLS - FLIGHT DATA COM | | | | MD-11, MD-11F | Installation of Wireless Quick Access RecorderThe existing Quick Access Recorder is removed and replaced with aTeledyneWireless Quick Access Recorder. | The Certification Basis for the original product as amended by thefollowingadditional or alternative airworthiness requirements:the following paragraph(s) at a later amendment:JAR 25 Amendment 16:§25.23, §25.25, §25.27, §25.561, §25.581(b)(1), §25.6 (...) | MOD-PCP-029, issue 02 (January 25, 2013) - Project Certification PlanMOD-MDL-029, issue 01 (June 14, 2013) - Master Data ListMOD-CCL-029, issue 01 (June 10, 2013) Compliance ChecklistMOD-EWA-029, issue 01 (December 11, 2012) - EWIS AssessmentMOD-1 (...) | Compatibility must be established for each aircraft. Compatibility has been established for the aircraft listed in MOD-MDL-029, issue 01 or later approved revision | 24.06.2013 |
| 10014159 | TELEDYNE LIMITED | EASA.A.S.01290 | | | BOEING 767-300 | Installation of Teledyne Controls Optical Quick Access recorder (OQAR) P/N 224800 | | Master Dat List STC-MDL-015-02 dated 27 September 2005 | | 26.10.2005 |
| 10014562 | TELEDYNE LIMITED | EASA.A.S.02662 | | | MD80 SERIES | Installation of Optical Quick Access Recorder. | | Master Data List MOD-MDL-018-01 dated 6 December 2006 or later EASA approved revision. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 05.07.2007 |
| 10039512 | TELEDYNE LIMITED | EASA.A.S.01198 | REV 1 | | 767-200, 767-300/-300F | Installation of Wireless Quick Access Recorder | The Certification Basis for the original product remains applicable tothis certificate/ approval.FAR-25The requirementsfor environmental protection and the associatedcertificatednoise and/ or emissions levels of the original product areunchanged and (...) | Teledyne Controls Master Data List STC-MDL-013-15 Issue 15,dated 04January 2012or later revisions of the above listed documents approved by EASA | None | 08.05.2012 |
| 10016095 | TELEDYNE TECHNOLOGIES, INC. | EASA.IM.A.S.02284 | REV. 2 | FAA STC ST01547LA | BD-700-1A10 | Installation of EMS Technologies High Speed Data Satellite Communicationsystem HSD-64 and HSD-128 and validation of FAASTC ST01547LA.This STC supersedes EASA.IM.A.S.02284 following a change in theapplicant's address. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | MDL: MDL246759 Rev. , dated August 1, 2003or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions ofthis decision) and/ or the Technical Implementation Procedures of EU/USA Bilateral Agreement. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce roadverse effect upon the airworthiness ofthe product. | 08.04.2013 |
| 10016174 | TELEDYNE TECHNOLOGIES, INC. | EASA.IM.A.S.02401 | REV. 1 | FAA STC ST00334LA | 747-400 | Installation of Aircraft Communications, Addressing and ReportingSystems (ACARS).This STC supersedes EASA.IM.A.S.02401 following a change in theapplicant's address. | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | AFM TDY-ACARS-747-ANA-1 Airplane Flight Manual Supplement, datedDecember 30, 1996Teledyne Controls Master Data List No.MDL 2231500-14C initial release,dated November 14, 1996or later revisions of the above listed documents approved by EASA inaccordra (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce roadverse effect upon the airworthiness ofthe product. | 29.01.2013 |
| 10027900 | TELEDYNE TECHNOLOGIES, INC. | | | FAA STC ST02172LA | B747-100, 747-200, 747-200B B747-200C, 747-200F, 747-300 B747-400, 747-400F | Installation of Teledyne Controls Wireless Ground Link - Quick AccessRecorder (WQAR) in accordance with FAA approved Teedyne Controls MasterData List Document No. MDL 2243800-747RRevision ".*" Appendix A, datedOctober 21, 2008 or later approved revisionsValidation of FAA STC ST02172LA | The Certification Basis for the original product remains applicable tothis certificate/ approval | or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions ofthis decision) | None | 13.11.2009 |
| 10031056 | TELEDYNE TECHNOLOGIES, INC. | | REV. 1 | FAA STC ST00498LA | B757-200 | Installation of Aircraft Communications, Addressing and Reporting System(ACARS) | The Certification Basis for the original product remains applicable tothis certificate.The requirements for environmental protection and the associatedcertificatednoise and/ or emissions levels of the original product areunchanged and remain applicable to this approval. | Teledyne Controls Master Data List No. MDL 2231500-820 Revision B,12/10/1998,Teledyne Controls AFM Supplement TDY-ACARS-757-COA-2 dated 30/09/1998, or later revisions of the abovelisted documents approved by EASA inaccordance with EASA EDDecision 20 (...) | None | 23.07.2010 |
| 10032418 | TELEDYNE TECHNOLOGIES, INC. | | REV. 2 | FAA STC ST02239LA | B777-200, B777-200LR, B777-300 B777-300ER, B777F | Installation of Wireless Ground Link - Quick Access Recorder (WQAR)This STC consists in the installation of the WQAR onBoeing 777 series. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval- EASA TODS IM.A.003- Special Condition H-01 'Enhanced Airworthiness Program for Aero (...) | Master Data List, Ref : MDL2243800-777, dated 03 March 2009Instruction for Continued Airworthiness for EWIS, Ref : EW2243800-B7-1Revision B, dated 16 June 2010or later revisions of the above listed documents approved by EASA inaccordance with EASA E (...) | This STC is applicable to the following WQAR p/n (2243800-xxx) :- 2243800-61- 2243800-62- 2243800-63- 2243800-64- 2243800-71- 2243800-73- 2243800-77- 2243800-79- 2243800-81- 2243800-360- 2243800-362- 2243800-364Prior to installation of this modiffica (...) | 02.05.2013 |
| 10033914 | TELEDYNE TECHNOLOGIES, INC. | P-EASA.IM.A.S.01234 | | FAA STC ST01778LA | A300B2-1A, -1C, -203 A300B2K-3C A300B4-2C, -103, -203 A300B4-601, -603, -605R A300B4-620, -622R A300C4-605 VARIANT F A300F4-605R, -622R A310-203, -204, -221, -222 A310-304, -322, -324, -325 | Installation of Wireless Ground Link - Quick Access Recorder (WQAR)This STC consists in the installation of the WQAR onAirbusA300/A310/A300-600 series. | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval- DGAC-F TCDS No 145The requirements for environmental protection and the associatedcer (...) | Master Data List, Ref : MDL2243800-31X, dated 14 March 2005Instruction for Continued Airworthiness, Ref : ICA2243800-A1-1, dated 23February 2005 or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 200 (...) | This STC is applicable to the following WQAR p/n (2243800-xxx) :- 2243800-21- 2243800-31- 2243800-33- 2243800-35- 2243800-37- 2243800-39- 2243800-42- 2243800-43- 2243800-44- 2243800-52- 2243800-53- 2243800-64- 2243800-71- 2243800-73- 2243800-75- 2243800-77- 2243800-79- 2243800-81- 2234572-001 | 18.02.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------|-------------------|----------------|----------------------|---|--|--|--|---|---------------|
| 10043150 | TELEDYNE TECHNOLOGIES, INC. | EASA.IM.A.S.0128 | REV. 1 | FAA STC ST01620LA | A318-111/-112/-121/-122 A319-115/-131/-132/-133 A320-111/-211/-212/-214 A320-215/-216 A320-231/-232/-233 A321-111/-112/-131 A321-211/-212/-213 A321-231/-232 | Installation of Teledyne Controls Wireless Groundlink QuickAccessReader (WQAR). This STC supersedes EASA.IM.A.S.0128 following a change in the applicant's address. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List ref. MDL2243800-32X, revision "-", dated March 8, 2004 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Technical Imp (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 14.02.2013 |
| 10043159 | TELEDYNE TECHNOLOGIES, INC. | EASA.IM.A.S.0092 | REV. 1 | ST01632LA | 757-200, 757-200PF, 757-300 | Installation of Teledyne Controls Wireless Groundlink - Quick Access Recorder in accordance with FAA ST01632LA. This STC supersedes EASA.IM.A.S.0092 following a change in the applicant's address. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List Drawing No MDL2243800-757, revision "-", dated 3 March 2004, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Technic (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 29.01.2013 |
| 10043160 | TELEDYNE TECHNOLOGIES, INC. | EASA.IM.A.S.00194 | REV. 2 | FAA STC ST01633LA | A330-201/A330-202/A330-203 A330-223/A330-243 A330-301/A330-321/A330-322 A330-323/A330-341/A330-342 A330-343 | Installation of a Wireless Groundlink Quick Access Recorder. This STC is a validation of the FAA STC ST01633LA. This STC supersedes EASA.IM.A.S.00194 following a change in the applicant's address. No additional technical investigation was performed. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Drawing List No. MDL2243800-330, rev., dated March 3, 2004 Teledyne Instructions for Continued Airworthiness Document No.ICA2243800-A3-1, dated October 27, 2003 or later revisions of the above listed documents approved by EASA in accordance with (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 08.04.2013 |
| 10043161 | TELEDYNE TECHNOLOGIES, INC. | | REV. 1 | FAA STC SA8030NM | 737-100, 737-200C 737-300, 737-400 737-500, 737-600 737-700, 737-800 | Requirement for the installation of Teledyne Controls Digital Flight Data Acquisition Unit (DFDAU) and Digital Flight Data Management Unit (DFDMU). This STC supersedes UK CAA's certificate CAA.21NE2.00007 following a change in the applicant's address. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List, MDL2233000-737 revision A, appendix C dated 26 June 2000 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Technical (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 08.04.2013 |
| 10043162 | TELEDYNE TECHNOLOGIES, INC. | 2004-5756 | REV. 2 | ST01268LA | 737-100, 737-200, 737-200C 737-300, 737-400, 737-500 737-600, 737-700, 737-800 737-900 | Installation of Teledyne Controls Wireless Ground Link-Quick Access Recorder (WGL-QAR). This STC supersedes Luftfahrt-Bundesamt (LBA) certificate LBA.21NE1.TA0743 and follows EASA approval letter 2004-5756 due to a change in the applicant's address. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Teledyne Controls MDL 2243800-737, Revision "B", Appendix A, Nov. 16, 2001 (>,-<)- or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Tec (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 15.04.2013 |
| 10043163 | TELEDYNE TECHNOLOGIES, INC. | | REV. 1 | FAA STC SA6084NM | DC-9-81 (MD-81) DC-9-82 (MD-82) DC-9-83 (MD-83) DC-9-87 (MD-87) MD-88 | Validation of STC Teledyne Controls Optical Quick Access Recorder (OQAR)/P/N 2248000-40 on McDonnell Douglas MD-80. This STC was previously approved by Registro Aeronautico Italiano according to their letter No.93/514/MAV and has been issued by EASA following a change in the applicant's address. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | FAA approved Teledyne Controls Master Data List No. MDL 2248000-MD-80, Revision "B", dated November 10, 1993 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 08.04.2013 |
| 10044644 | TELEDYNE TECHNOLOGIES, INC. | | | FAA STC ST1805LA | B767-200/-300/-300F/-400ER | Installation of Wireless Ground Link - Quick Access Recorder (WQAR) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Teledyne Controls Master Data List, MDL 2243800-767, Revision A, 29/04/2005 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Technica (...) | None | 26.04.2013 |
| 10014312 | TENENCIA Ltd. | EASA.A.S.01557 | REV. 1 | | 1. BEECH 58PA 2. BEECH 58/58A 3. BEECH G58 | Hawker Beechcraft Baron Steep Approach Landing Certification This Certification at Revision 1 of the STC ref. EASA.A.S.01557 includes additional aeroplane models compared to the initial issue of the STC | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Tenencia Ltd Modification TEN 426, "Hawker Beechcraft Baron Steep Approach Landing Certification", Issue 4 dated 20 May 2010. Tenencia Ltd AFM Supplement 1 Issue 2 to the Beech 58PA Pilot's Operating Handbook and FAA Approved Airplane Flight Manual P/N10 (...) | Hawker Beechcraft Model 58PA S/N TJ-256 and Hawker Beechcraft Model 58 S/N TH-1967 only. | 26.05.2010 |
| 10014396 | TENENCIA Ltd. | EASA.A.S.01690 | | | BAE JETSTREAM 3100 | Installation of Dual Mode S Transponders & Honeywell CAS 67A TCAS II. | | DD/AA/397 AAE Supplement 4 Issue 1 or later approved revision to Approved AFM Ref. HP4.11 | | 04.01.2006 |
| 10014498 | TENENCIA Ltd. | EASA.A.S.02561 | | | BAE ATP | Installation of No. 2 ECS Pack | | Tenencia Modification TEN 460 EASA approved Tenencia AFM Supplement 2 Issue 1 or later approved revision. | | 23.10.2006 |
| 10014514 | TENENCIA Ltd. | EASA.A.S.02589 | | | LOCKHEED 188 | Installation of Mode S Transponders for Enhanced Surveillance | | Tenencia Modification TEN 462 Tenencia AFM Supplement 12 Issue 1 dated May 2006 or later approved revision | None | 11.05.2006 |
| 10014583 | TENENCIA Ltd. | EASA.A.S.02689 | | | BAE 146 BAE 146/AVRO 146 SERIES | Installation of Single or Dual Universal UNS-1L or UNS-1K+ Flight Management System | | Tenencia Modification TEN 467 AFM Supplement 1 issue 1 (TEN/146/001) AFM Supplement 2 issue 1 (TEN/146/002) AFM Supplement 3 issue 1 (TEN/146/003) AFM Supplement 4 issue 1 (TEN/146/004) | RNAV Approaches are not permitted for installation with electronic-mechanical instruments | 06.09.2006 |
| 10014710 | TENENCIA Ltd. | EASA.A.S.02882 | | | BAE 146/AVRO 146 SERIES BAE 146/RJ SERIES | Installation of Cockpit Door Locking System. The modification to the Cockpit Door provides the ability for the Cabin Crew to access the Cockpit in the case of Pilot incapacitation. | | - Tenencia Modification Ten 487 Issue 1 - Tenencia AFM Supplement 5 Issue 1 to the Bae 146/AVRO RJ Flight Manuals ref. TEN/146/005. | BAE Systems Modification HCM50303C. | 12.12.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------|------------------|----------------|----------------------|--|---|--|---|--|---------------|
| 10014746 | TENENCIA Ltd. | EASA.A.S.02929 | | | BAE ATP | Lead Acid Battery Installation. | | - TEN 494 Issue 4- Report TEN 494/CR Issue 2- Report TEN 494/1 Compliance Report Issue 1 | 1. Applicable only to aeroplanes that are fitted with an appropriately approved Class E Cargo Compartment used only for the carriage of cargo.2. Prior to installation of this modification the installer must determine that the interrelationship (...) | 27.01.2009 |
| 10014758 | TENENCIA Ltd. | EASA.A.S.02947 | | | AIRBUS A300 B4 AIRBUS A300 MODEL B2 | Installation of dual Universal UNS-1L or UNS-1K+ Flight Management System (FMS) This modification allows complying with Precision RNAV operations in designated European airspace. | | Certification Report, ref : TEN 495 Issue 3 Engineering Order, ref ECS 204775 revision Dor any later approved revisions of these documents Aircraft Flight Manual Supplement, ref : TEN/A300/001 issue 3 Instruction for Continued Airworthiness, ref : ECS 205511 revision N/C | The installation of this STC has a prerequisite aircraft configuration. This STC is only applicable on aircraft :1. in which the following modifications have been embodied :- FAA STC ST01113CH, accepted by EASA under the grandfather rights ruling (...) | 22.02.2008 |
| 10014765 | TENENCIA Ltd. | EASA.A.S.02955 | | | BOEING 737 | Installation of Mode S Transponders for Enhanced Surveillance. | | Tenencia Report TEN 500, Issue 2, dated 4/05/07# Tenencia Report TEN 500/1, Issue 2, dated 4/05/07# Flight Manual Supplement TEN/B737/001, Issue 1 | - Limited to installation on Boeing 737-82R S/N 29329 only. - The Operators Minimum Equipment List (MEL) should be revised to include the requirements for Mode S Transponders and Elementary and Enhanced Surveillance in accordance with Boeing Aircraft Company Master Minimum Equipment List (M MEL) | 31.05.2007 |
| 10014771 | TENENCIA Ltd. | EASA.A.S.02963 | | | BOEING 757-200 | Installation of Mode S Transponders for Enhanced Surveillance | | Installation of Mode S Transponders (ref: TEN 501, Issue 1 20/03/2007) AFM Supplement (TEN/B757/001, Issue 1) | Certification Basis as per TCDS including: JAA TGL 13, revision 1, 01/06/2003 EASA AMC 20-13, 22/12/2006 | 03.04.2007 |
| 10015039 | TENENCIA Ltd. | EASA.A.S.03352 | | | AVRO RJ | VIP Interior Installation. | | Modification TEN 532, Bae 146 Series VIP Installation, Issue 1 dated 14 May 2008 Tenencia Ltd. Report TEN 532/1.1 Certification Compliance Report, VIP Configuration Installation, Issue 1 dated 28 April 2008, Tenencia Ltd. | S/N E3386 only. | 19.05.2008 |
| 10015160 | TENENCIA Ltd. | EASA.A.S.03529 | REV. 1 | | AVRO 146-RJ70 | VIP Installation | | 1. Modification TEN 549.2, Bae 146 Series VIP Installation Issue 1 dated 28 October 2008. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 19.06.2009 |
| 10015251 | TENENCIA Ltd. | EASA.A.S.03696 | | | EMBRAER EMB-135 SERIES | EMB 135 Electronic Flight Bag installation | | TEN 582 Issue 2 dtd. 13.5.09 *) - Flight Manual Supplement TEN/EMB135/001 Issue 2 *) or later approved revision | None | 29.05.2009 |
| 10015265 | TENENCIA Ltd. | EASA.A.S.03727 | | | BOEING 747-400 SERIES | Installation of 3 AD Aerospace FV-0315 External Camera's and 1 AD Aerospace FV-0610 Video Transmission Unit in accordance with Tenencia Ltd modification TEN 587 Issue 1, or later approved revisions. | | - TEN 587/4 Iss. 1 Maintenance Manual Supplement - TEN 587/7 Iss. 1 Operations Manual Supplement | 1. Tenencia Ltd. Instruction for Continued Airworthiness, Report TEN 587/4, issue 1, shall be incorporated in the aircraft maintenance documentation. 2. Tenencia Ltd Operations Manual Supplement, report TEN 587/7 Iss. 1, shall be incorporated (...) | 16.04.2009 |
| 10015269 | TENENCIA Ltd. | EASA.A.S.03741 | | | BOEING 757 | 78 Seat Installation / This modification consists of: - Removal of existing 200 Economy Seats. - Installation of 78 First Class Seats including 3 Club 4 installations, 2 at the forward and 1 at the aft end of the cabin. | | TEN 591 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 30.03.2009 |
| 10017107 | TENENCIA Ltd. | EASA.R.S.01082 | | | AS 355 F, AS 355 N | Installation of a Gyrone 935 Camera mounted on the centerline of the aircraft in front of the nose | | Tenencia modification report: TEN 428 | | 05.04.2006 |
| 10017207 | TENENCIA Ltd. | EASA.R.S.01359 | | | EUROCOPTER FRANCE AS 350 SERIE | Gyrone 935 camera installation on the AS 350 series a/c. | | Installation according to report n. TEN 428 Flight Manual Supplement TEN/EUROCOPTERAS350/FM/SUPP/001M aintenance Manual Supplement TEN/EUROCOPTER/MM/SUPP/001 | | 06.03.2007 |
| 10028847 | TENENCIA Ltd. | P-EASA.A.S.03872 | REV. 2 | | AVRO 146-RJ100 | Avro 146-RJ100 Amiri Flight VIP Installation. | EASA TCDS A.182 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-01 Instructions for Continued Airworthiness (ICA) on Electrical Wi (...) | Modification TEN 621, BAE Systems Avro 146-RJ100 Amiri Flight VIP Installation, Issue 5 dated 24 January 2011 or later revisions of the above listed documents approved by EASA | Serial number E3387 only. | 26.01.2011 |
| 10032387 | TENENCIA Ltd. | | | | MK050, MK0502 | Installation of a Universal Avionics Terrain Awareness and Warning System (TAWs) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Electronic Wiring Interconnection System (EWIS) as defined in CS 25.170 in accordance (...) | Certification Plan CP/TEN/645 Issue 1- Ten supplement 1 issue 1 to the Fokker F27 Mk 0502 flight manual or later revisions of the above listed documents approved by EASA | As contained in Ten supplement 1 issue 1 to the Fokker F27 Mk 0502 flight manual | 02.11.2010 |
| 10032536 | TENENCIA Ltd. | P-EASA.A.S.03438 | | | BOEING 747-200 BOEING 747-300 | Installation of a Dual Honeywell HT-9100 FMS per Modification Description TEN 539 Boeing 747-200 Dual Honeywell HT9100 GNSS Navigation Management System Installation, Issue 4 dated 10 November 2010 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval FAR Part 25 Amendment 39 JAA Guidance Material TGL 10, November 1, 2000 Airworthiness an (...) | AFM supplement TEN SUPPLEMENT 1 ISSUE 3 for the Boeing 747 series Flight Manual Report TEN 539 /7 Issue 2, Operations Manual Supplement Boeing 747HT9100 FMS Installation or later revisions of the above listed documents approved by EASA | This STC is applicable to aircraft that have previously been modified in accordance with STC ST01841CH / I.M.A.S.01660 GPS Antenna Provision Installation and have dual Honeywell Rot Computers P/N 2590623-914 or-915 | 12.11.2010 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------|------------------|----------------|----------------------|---|--|---|---|---|---------------|
| 10033233 | TENENCIA Ltd. | | | | AVRO 146-RJ85 | Modification TEN 655, AVRO 146-RJ85, Installation of 3R and 4R Galleys and No. 3 VHF COMM | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:JAR-25 at Change 10 in lieu of JAR-25 at Change 5,Electronic Wiring Interconnection Sys (...) | Modification TEN 655, AVRO 146-RJ85, 3R and 4R Galley and No 3 VHF CommInstallation. Tenencia Ltd document number TEN 655. Issue 3 dated 24th December 2010 or later revisions of the above listed documents approved by EASA | Serial numbers E2323 and E2325 only. | 05.01.2011 |
| 10033789 | TENENCIA Ltd. | | | | F27 MK 050 & F27 MK 0502 | Installation of a Single UNS-1Lw Universal Avionics Flight Management System. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification Plan CP/TEN/644 issue 1 Installation report and Statement of Compliance DF/ENG/014 Issue 7 System Safety Assessment 387-01-0008 Revision B Master Drawing List 288-00-0001 of 31 March 2009 AFM Supplement TEN/FOKKER F27/002 Issue 1 Additional (...) | AFM Supplement TEN/FOKKER F27/002 Issue 1 | 08.02.2011 |
| 10034446 | TENENCIA Ltd. | | | | B737-300, -400, -500, -600 B737-700, -800, -900 | Cockpit Door Surveillance System | Certification Basis as per EASA TCDS IM.A.080 Issue 3, 26th October 2010 plus: The applicant elected to comply with JAR 25 at change 15 for all models. In addition, CRI H1 issue 2 dated 23/02/2011 on 'Enhanced Airworthiness Programme for Aeroplane System (...) | - Modification TEN 660 B737 Series Cockpit Door Surveillance System issue 2 dated 25 February 2011 - Report TEN 660/1 Certification Compliance Report B737 Series Cockpit Door Surveillance System issue 1 dated 4 February 2011 - Certification Plan CP/TEN/ (...) | Not applicable | 05.04.2011 |
| 10034998 | TENENCIA Ltd. | | | | A318, A319, A320, A321 | 1 NET In-flight Entertainment (IFE) System Installation | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval 1. CRI H-01 'Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS' The re (...) | - Modification TEN 646, ref report TEN 646, Issue 1, dated 17th May 2011 - Certification Compliance Report, ref. report TEN 646/1, Issue 1, dated 6th May 2011 - Electrical Wiring Interconnection System (EWIS) ICA, ref. report 646/26A, Issue 1, dated 11th (...) | Not applicable | 18.05.2011 |
| 10035047 | TENENCIA Ltd. | | | | A319, A320, A321 ALL MODELS | 'Installation of provisions for the use of a Class 2 Dual Goodrich/Electronic Flight Bag (EFB) Smart Display System' | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval 1. CRI H-01 'Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS' The re (...) | - Modification TEN 674, ref. report TEN 674, Issue 2, dated 10th May 2011 - Certification Compliance Report, ref. report TEN 674/1, Issue 2, dated 10th May 2011 - A319/A320/A321 Airplane Flight Manual Supplement, ref. Doc.TEN/A320/002, Issue 2 or later re (...) | Not Applicable | 23.05.2011 |
| 10035689 | TENENCIA Ltd. | | | | BAE 146 SERIES 200 | EWIS Assessment for VIP Cabin Interior | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval Electronic Wiring Interconnection System (EWIS) requirements in CS-25 Amendment 8 Appen (...) | Modification TEN 664, EWIS Assessment for VIP Cabin Interior EASA STC10027954. Tenencia Ltd document number TEN 664, Issue 1 dated 25th January 2011. Aviation Traders Limited VIP Cabin Interior Modification Summary Sheet Issue 2 for MOD-11193-00001 or (...) | Serial number E2079 only. | 12.07.2011 |
| 10037413 | TENENCIA Ltd. | P-EASA.A.S.03871 | | | A300-600 SERIES | Cockpit Door Monitoring System Installation of the AD Aerospace "Cabin Vu" cockpit door monitoring system (video monitor, monitor control panel, video switch unit and three cameras) in accordance with Tenencia Ltd Modification Report (Master Data List) T (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval CRI H-01 Special Condition for Enhanced Airworthiness Programme for Aeroplane system - 1 (...) | Tenencia Maintenance Manual Supplement TEN 656/4 issue 1 Tenencia Operations Manual Supplement TEN 656/7 issue 1 Tenencia Installation Instructions (Configuration 1) TEN 656/10. 1 issue 1 Tenencia Illustrated Parts Catalog TEN 656/17 issue 1 Tenencia Weigh (...) | None | 25.11.2011 |
| 10039138 | TENENCIA Ltd. | | | | A319-111/-112/-113/-114/-115 A319-131/-132/-133 A320-111/-211/-212/-214/-215 A320-216/-231/-232/-233 A321-111/-112/-131/-211/-212 A321-213/-231/-232 | In-Seat Power Supplies | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Modification TEN 651 Issue 1 Flight Manual Supplement TEN/A320/003 Issue 1 or later revisions of the above listed documents approved by EASA | None | 12.04.2012 |
| 10040700 | TENENCIA Ltd. | | | | AVRO 146 SERIES RJ100 AVRO 146 SERIES RJ115 AVRO 146 SERIES RJ70 AVRO 146 SERIES RJ85 BAE 146 SERIES 100 BAE 146 SERIES 200 BAE 146 SERIES 300 | ADS-B Out Surveillance | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Modification TEN 706 Issue 1 dated 13 June 2012. Flight Manual Supplement TEN/146/016 Issue 1 or later revisions of the above listed documents approved by EASA | Configuration using transponder part number 751-7800-11005 does not meet EASA AMC 20-24 for ADS-B NRA. Emergency Status/Emergency Indicator are not provided. | 20.07.2012 |
| 10043083 | TENENCIA Ltd. | EASA.A.S.03812 | | | 737-300 | Collins VHF-700B Installation (8.33 kHz) | | Tenencia Documents: - Modification TEN 596 Issue 2 dated 11 May 2009 - TEN 596/4 Maintenance Manual Supplement | Limited to Boeing 737-300; JY-JAX, s/n 23955; JY-JAN, s/n 23956. | 12.05.2009 |
| 10043347 | TENENCIA Ltd. | | | | A310-304 | Avionics and Cabin Systems Upgrade | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI D-01, Glass Screens of Displays/Monitors CRI H-01, Enhanced Airworthiness Programme for Aeroplane Systems #CA on EWISS (...) | Modification TEN 687, Airbus A310-304 Avionics and Cabin Systems Upgrade, Issue 2 dated 17 January 2013 Report TEN 687/1, Certification Compliance Report, Avionics and Cabin Systems Upgrade, Issue 2 dated 17 January 2013 or later revisions of the above listed documents approved by EASA. | Aircraft Serial Number 431 only. | 22.01.2013 |
| 10044218 | TENENCIA Ltd. | | | | ISLANDER, BN2A-8, BN2A-26 | Aerial Spray Installation | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - the following paragraph(s) at a later amendment: Applicable paragraphs of BCAR Section D Issue 14 and BCAR Section K Issu (...) | Modification TEN 718, BN2A Aerial Spray Installation, Report TEN 718, Issue 3 dated 25 March 2013. Maintenance Manual Supplement, Islander BN2A, BN2 Spray System, Report TEN 718/4, Issue 2 dated 23 March 2013. Flight Manual Supplement, Aerial Spray Inst (...) | Only non flammable and non toxic detergents are to be used in this installation. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and (...) | 26.03.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------|-------------------|----------------|----------------------|---|---|--|---|---|---------------|
| 10045481 | TENENCIA Ltd. | | | | A319-133 | Galley Complex Installation.Installation of Galleys G1 and G2 and a Sliding Door. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI D-01 Installation of Interior Doors The requirements for environmental protection and the associated certified noise an (...) | Modification TEN 722, Airbus A319-133, Galley Complex Installation.Issue 1 dated 10 June 2013 EWIS ICA, TEN 722/26A Issue for later revisions of the above listed documents approved by EASA. | Airbus A319-133 serial number 4842 only.This Galley Complex Installation is part of another EASA STC for this aircraft, Interior Outfitting Installation, EASA Project Number 010014342-001 and EASA STC Approval 10046155 refer. This Galley Complex Install (...) | 23.08.2013 |
| 10046600 | TENENCIA Ltd. | | | | A319-111/-112/-113/-114/-115 A319-131-132/-133 A320-111/-211/-212/-214/-215 A320-216/-231/-232/-233 A321-111/-112/-131/-211/-212 A321-213/-231/-232 | Installation of Pitch Aircraft Seating System Models QCC1300 and QCC1200 Seats | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Certification Plan CP/TEN/689 Issue 4, dated May 29, 2013 Certification Compliance Report TEN 689/1 Issue 1, dated 18 September 2013 Modification Report and Declaration of Compliance TEN 689 Issue 1, dated 18 September 2013 Instructions for Continued Airw (...) | None | 02.10.2013 |
| 10016738 | TEST INSTRUMENTATION INC. | EASA.IM.R.S.01063 | | FAA STC SR00052SE | R44, R44 11 | Installation of Classic Aircraft products Cargo Hook System CH-800 in accordance with FAA STC SR00052SE | | EASA approved Rotorcraft Flight Manual No 44-510 EU | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 21.04.2006 |
| 10045993 | TESTORI AERO SUPPLY S.R.L. | | | | 777-200ER | Y/C CUSHION FOAMS & FABRIC SEAT COVERS KIT [Replacement of Y/C CUSHION FOAMS & FABRIC SEAT COVERS with new ones designed and manufactured by Testori Aero Supply (TAS) ON B777 A/C equipped with Recaro Seats.] | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | ...MASTER DOCUMENT LIST DOC. TAS-UN-STC-MDL-01 REV. 2 DATED 25/07/2013 ...MASTER DRAWING LIST DOC. TAS-UN-STC-MWL-01 REV.0 DATED 12/05/2013 or later revisions of the above listed documents approved by EASA. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 02.08.2013 |
| 10015423 | TEXAS SKYWAYS, INC. | EASA.IM.A.S.01154 | | FAA STC SA09133SC | CESSNA 182 CESSNA 182A CESSNA 182B CESSNA 182C CESSNA 182D CESSNA 182E CESSNA 182F CESSNA 182G CESSNA 182H CESSNA 182J CESSNA 182K CESSNA 182L CESSNA 182M CESSNA 182N CESSNA 182P CESSNA 182Q CESSNA 182R CESSNA F182-P CESSNA F182-Q | Installation of Teledyne Continental Motors IO-550-F engine and Hartzell or McCauley Propeller in acc. with Texas Skyways Spec. TS0028, Rev. 1, or later approved Revision and TS0034 | | FAA STC SA09133SC Texas Skyways AFM-S FS550182, Rev. 2, or later approved Revision Texas Skyways TachOpMan TS0037, dated 27 March 1995, or later approved Revision | The approval is only valid in combination with the installation of STC EASA.A.S.01374 (Gomolzig GmbH, Germany) | 07.07.2006 |
| 10015424 | TEXAS SKYWAYS, INC. | EASA.IM.A.S.01155 | | FAA STC SA10070SC | CESSNA 180 CESSNA 180A CESSNA 180B CESSNA 180C CESSNA 180D CESSNA 180E CESSNA 180F CESSNA 180H CESSNA 180J CESSNA 180K CESSNA 182 CESSNA 182A CESSNA 182B CESSNA 182D CESSNA 182E CESSNA 182F CESSNA 182G CESSNA 182H CESSNA 182J CESSNA 182K CESSNA 182L CESSNA 182M CESSNA 182N CESSNA 182P CESSNA 182Q CESSNA 182R CESSNA 185 CESSNA 185A CESSNA 185B CESSNA 185C CESSNA 185D CESSNA 185E | Installation of Hartzell 3-blade Propeller in acc. with Texas Skyways Doc. 5016-0000, Rev. E, or later approved Revision | | FAA STC SA10070SC Texas Skyways AFM-S 5016-180, Rev. IR, or later approved Revision for Cessna 180 series Texas Skyways AFM-S 5016-182, Rev. IR, or later approved Revision for Cessna/ Reims 182 series Texas Skyways AFM-S 5016-185, Rev. IR, or later appr (...) | The approval is only valid with latest EASA approved Noise Certificate | 07.07.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------|---------------------|----------------|-------------------------------|--|--|---|--|--|---------------|
| 10015425 | TEXAS SKYWAYS, INC. | EASA, I.M.A.S.01156 | | FAA STC SA09964SC | CESSNA 180 CESSNA 180A CESSNA 180B CESSNA 180C CESSNA 180D CESSNA 180E CESSNA 180F CESSNA 180G CESSNA 180H CESSNA 180J CESSNA 180K CESSNA 182 CESSNA 182A CESSNA 182B CESSNA 182C CESSNA 182D CESSNA 182E CESSNA 182F CESSNA 182G CESSNA 182H CESSNA 182J CESSNA 182K CESSNA 182L CESSNA 182M CESSNA 182N CESSNA 182P CESSNA 182Q CESSNA 182R CESSNA F182-P CESSNA F182-Q | Installation of engine mount heat shields in acc. with Texas Skyways Doc. 5018-101, dated 12 Oct 2001, or later approved Revision | | FAA STC SA09964SC | | 07.07.2006 |
| 10016688 | TEXAS SKYWAYS, INC. | EASA, I.M.E.S.01000 | | FAA STC SE09131SC | | Removal of fuel injection system and induction system and installation of carburetor in accordance with Texas Skyways, Inc. Specification TS0030, dated 01 May 1995 or FAA later approved revision, based on FAA Supplemental Type Certificate Number SE09131SC | | Texas Skyways, Inc. Specification TS0030, dated 01 May 1995 or later approved revision. | Approval of this change in type design applies to the above-mentioned model only. Compatibility of this modification with previously approved modifications must be determined by the installer. A copy of this Certificate must be maintained as part of the (...) | 06.04.2006 |
| 10038267 | THALES AVIONICS Inc. | | REV. 1 | FAA STC NO. ST02411LA | 777-200 | In-Flight Entertainment (IFE) System Installation | IM.A.003The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CRI H-01 EWISThe requirements for environmental protection and the associated de (...) | FAA STC No. ST02411LA, dated 20 January 2012. Thales Avionics Index List, Drawing No. 254709, Revision "A", dated December 16, 2011, or later approved revision. The instructions for Continued Airworthiness is called out in its associated Master Drawing (...) | Concurrent installation of the following STCs are required for their installation of this STC: 1) Flight Structures, Inc. EASA STC No. 10015720, Business Class Seat Installation, 2) Northwest Aerospace Technologies, Inc. EASA STC No. 10038268 (EASAProject (...)) | 03.02.2012 |
| 10041205 | THALES AVIONICS Inc. | | | FAA STC ST02431LA | A321-211, A321-212, A321-213, A321-231, A321-232 | Installation of Thales Avionics IFS TopSeries In-Flight Entertainment (IFE) System | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - FAA Supplemental Type Certificate ST02431LA issued 9th August 2012- PSCP-2011-1001 Rev.F, dated 9th July 2012- Index List 255058 Rev.B, dated 8th August 2012- Reference Data List 500440-103RDL Rev.J, dated 18th July 2012- Master Drawing List 50044 (...) | This STC must be installed in conjunction with EASA STC 10040704 - Aerospace Engineering Solutions (AES) - Introduction of Twin Class Cabin Configuration | 29.08.2012 |
| 10042547 | THALES AVIONICS Inc. | | | | A321-211, -212, -213, -231, -232 | Installation of Thales Avionics IFS TopSeries In-Flight Entertainment (IFE) System. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - FAA Supplemental Type Certificate ST02435LA issued 21st November 2012- PSCP-2011-1003 Rev. H, dated 10th August 2012- Index List 255193 Rev. A, dated 21st November 2012- Reference Data List 500440-101RDL rev. L, dated 21st November 2012- Master Draw (...) | This STC must be installed in conjunction with EASA STC 10040704 - Aerospace Engineering Solutions (AES) - Introduction of Twin Class Cabin Configuration. | 07.12.2012 |
| 10042295 | THE BOEING COMPANY | EASA, I.M.A.S.03158 | REV. 1 | FAA STC ST01697SE | 737-700/800/900ER | Lower Cabin Altitude (LCA) modification: cabin altitude reversion to 6,500 ft in lieu of 8,000 ft at 41,000 ft flight altitude. Modification in accordance with Boeing's Master Document List for Lower Cabin Altitude modification D926A200 Lower Cabin A (...) | B737-700/800. The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. B737-900ER: The applicable certification basis is detailed in the annex to this certificate. ANNEX The applicable Certification Basis f (...) | - "Master Drawing List"; Figure 1 of #Master Document List for Lower Cabin Altitude Modification, Boeing Model 737 Airplanes# D926A200 revision AM dated September 17, 2012-737-700 FAA Approved Airplane Flight Manual" D631A001 unique to each BBJ aeropl (...) | - Approval of this change in type design is limited to the eligible airplane serial numbers listed in Figure 1 of D926A200- It is the Operator's responsibility to ensure that all future repairs, alterations and modifications (RAM) affecting the pressur (...) | 04.07.2013 |
| 10016326 | THE BOEING COMPANY - | EASA, I.M.A.S.02635 | | FAA STC SA2123CE-D AMENDED 19 | BOEING 747-285B | Conversion of Passenger Airplane to dedicated Special Freighter with Main Deck Side Cargo Door in accordance with Boeing Drawing List document D3-8857, Section V, Appendix 31, Rev. CN | | D6-13703.3B5 Airplane Flight Manual Supplement D6-30680-34 B747-200 Cargo Loading Manual | This approval is limited to airplane model B747-200B S/N 22482 and S/N 22485. Boeing Commercial Airplane Group - Wichita Division, Airplane Flight Manual Supplement dated 9 September 1991 or revision for AFM D6-13703.3B5 or later approved revisions o (...) | 28.01.2008 |
| 10016519 | THE BOEING COMPANY - | EASA, I.M.A.S.02891 | | | | Conversion of Passenger Airplane to Main Deck Side Cargo Door Dedicated Special Freighter as defined by FAA approved Boeing Document D800-10610-7, Revision New or later FAA approved revisions. (Validation of FAA STC ST00252W1-D) | | FAA Approved Airplane Flight Manual Supplement to Airplane Flight Manual D6-35747 for Boeing Model 747-200B D6-30680 B747-200 Cargo Loading Manual. | FAA Approved Airplane Flight Manual Supplement to Airplane Flight Manual D6-35747 for Boeing Model 747-200B is required to support this modification of applicable airplanes as identified in table 1 of Boeing Doc. D800-10610-7, Rev. New or later FAA a (...) | 06.10.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|-------------------|----------------|--------------------------------|--|---|--|--|--|---------------|
| 10016644 | THE BOEING COMPANY - | EASA.IM.A.S.03085 | REV. 1 | FAA STC SA2124CE-D | BOEING 747-200B | Conversion of Passenger Airplane to Main Deck Side Cargo Door Dedicated Special Freighter in accordance with Boeing Drawing List document D3-8857, Section V, Appendix 33, Rev. CS. | | Boeing Commercial Airplane Group - Wichita Division, Airplane Flight Manual Supplement dated 9 September 1992 or later revision approved in accordance with ED Decision 2004/04/CF dated 10 December 2004.D6-30680-7 Cargo Loading Manual | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. Th (...) | 09.02.2009 |
| 10015813 | THE GOODYEAR TIRE & RUBBER COMPANY | EASA.IM.A.S.01882 | | FAA STC ST03183AT | 1. AIRBUS A330 2. AIRBUS A340 | Installation of Goodyear Radial Tire Size 1050x395R16, P/N 109Q89 1 on Airbus A330 and A340 Nose Landing Gear according to FAA STC ST03183AT | | - FAA STC ST03183AT, dated 26 October 2005,- Goodyear SB, Ref. 2005-32-002 dated 8 June 2005,- Goodyear SB, Ref. 2005-32-003 dated 8 June 2005 | | 19.09.2006 |
| 10015963 | THE NORDAM GROUP | EASA.IM.A.S.02112 | | FAA STC ST10526SC | LEARJET 45 | Replacement of Thrust Reverser Control Unit FAA STC ST10526SC | | FAR Compliance Matrix NTR-ER 0403, rev. CJAR Compliance Matrix NTR-ER 0905, rev. A Master Drawing List NTR-ER 0405, rev. E Instructions for Continued Airworthiness NTR-ER 0406, rev. A | | 21.03.2007 |
| 10016926 | THE NORDAM GROUP | EASA.IM.R.S.01312 | | FAA STC SH2554NM / EASA STC 20 | 369D/E/FF/H/HM/H/E, 500N, 600N | Installation of modified front window canopy. Note: This STC supersedes EASA STC.2004-112. | | Platinum Aviation Group Inc Master Drawing List TV 500-600 Revision A dated February 24, 2001 & Platinum Aviation Group Inc. Installation Instructions TV 500-600 True Vision Kit Revision Original dated April 21, 2000 or later approved revision. | | 24.01.2007 |
| 10038928 | THE NORDAM GROUP, Inc. | | REV. 1 | FAA STC ST09034SC | A319, A320, A321 | Installation of C & X Band Fuselage Nose Radome Shell | In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. SC-H01 Enhanced Airworthiness Programme for Aeroplane S (...) | - FAA STC ST09034SC Master Data List, ref. doc. MDL-190-2460-2, Rev. E, dated 29th July 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 27.03.2012 |
| 10014219 | THIELERT AIRCRAFT ENGINES GmbH | EASA.A.S.01380 | REV. 5 | | DR 400 RP, DR 400/120 D DR 400/140 B DR 400/180 R, DR 400/200 R | 1. Installation of TAE 125 engine. 2. Installation of TAE 125-02-99 engine and installation of FADEC Backup Battery. 3. Installation of TAE 125-02-114 engine. Revision 5 follows a changed ownership. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. JAR 23, Amdt. 1 Initial Issue (not required by 21A.101, elected to comply by the applic (...) | For TAE 125 engine: 1. Type Design Definition TD 60-012, Pilot's Operating Handbook for DR-400/120D/140B/180R/200R equipped with TAE 125. Supplemental Airplane Maintenance Manual AMM-60-01 For TAE 125-02-99 engine: 1. Type Design Definition TDD 60-01 Is (...) | Installation of TAE 125-02-114 is limited to DR 400 models with S/N 2574 and upwards. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification a (...) | 19.08.2013 |
| 10014287 | THIELERT AIRCRAFT ENGINES GmbH | EASA.A.S.01527 | REV. 7 | LBA EMZ SA1295 | 172F, 172G, 172H, 172L, 172K, 172L, 172M, 172N, 172P, 172R, 172S F 172F, F 172G, F 172H F 172K, F 172L, F 172M, F 172N F 172P | Major Change to STC EASA.A.S.01527, Revision 5: 1. Installation of engine TAE 125-01 in accordance with Installation Manual IM-20-01 and Type Design Definition Doc. No: TDD-20-01.2. Installation of engine TAE 125-02-99 in accordance with Installation (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. JAR 23, Amendment 1 (not required by 21A.101, elected to comply by applicant). This cer (...) | Instructions for operation: Cessna (F) 172-N, P with TAE 125-01 installation: Anhang Flughandbuch für (Reims) Cessna (F) 172 N & P mit TAE 125 Installation, TAEN: 20-0310-20011, Ausgabe 1, LBA anerkannt am 31.10.2002 oder jede spätere LBA/EASA-amerkan (...) | This STC supersedes LBA-EMZ SA 1295 Power/Rotational speed TAE 125-01 & 125-02-99: Take off power: 99 kw at 2300 min ⁻¹ (Propellerspeed) Max. continuous power: 99 kw at 2300 min ⁻¹ (Propellerspeed) Fuel: Service Bulletin TM TAE 601-0005 MTOW: Model C (...) | 19.08.2013 |
| 10014364 | THIELERT AIRCRAFT ENGINES GmbH | EASA.A.S.01632 | REV. 9 | LBA-EMZ SA 1377 | PA-28-140, PA-28-150 PA-28-151, PA-28-160 PA-28-161, PA-28-180 | 1. Installation of engine TAE 125-01 in accordance with Installation Manual IM-40-01 and Type Design Definition Doc. No.: TDD 40-01.2. Installation of engine TAE 125-02 in accordance with Installation Manual IM-40-02 and Type Design Definition Doc. No (...) | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. JAR 23, Amendment 1 (not required by 21A.101, elected to comply by the applicant) This c (...) | Instructions for operation: 1. For PA-28-161 with TAE 125-01 installation: Anhang Flughandbuch für Piper PA-28-161 mit TAE 125 Installation: Ausgabe 1, Rev. 0 Approved March 8, 2004 or each later EASA approved issue (see TM TAE000-0004). Supplement Pilot' (...) | 1. This STC supersedes LBA-EMZ SA 1377.2. Power/Rotational Speed: For TAE 125-01 and 125-02-99 installations: Take off power: 99 Kw at 2300 min ⁻¹ (Propellerspeed) Max. continuous power: 99 Kw at 2300 min ⁻¹ (Propellerspeed) For TAE 125-02-114 installation (...) | 19.08.2013 |
| 10014442 | THIELERT AIRCRAFT ENGINES GmbH | EASA.A.S.02492 | REV. 1 | | SR22 | Installation of engine Centurion 4.0 in accordance to Type Design Definition Doc. No.: TDD-70-01 Engine: Centurion 4.0 Propeller: MTV-14-D/195-30b Certification Basis: see CRI A-1 max. Take Off Power: 228 kw at 2300 min ⁻¹ (Propeller-RPM) max. (...) | | - Supplement Pilot's Operating Handbook for the Cirrus SR 22 equipped with TAE Centurion 4.0 installation; Doc. No.: 70-0310-70021, issue 1 or later approved issue- Supplement Maintenance Manual Cirrus SR 22 / Centurion 4.0, Doc. No.: AMM-70-01 (...) | | 19.08.2013 |
| 10014452 | THIELERT AIRCRAFT ENGINES GmbH | EASA.A.S.02505 | | | CESSNA 172 | Installation of Universal Fuel System | | | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 07.02.2006 |
| 10014500 | THIELERT AIRCRAFT ENGINES GmbH | EASA.A.S.02565 | REV. 3 | | U206F, TU206G U206B, TU206F U206H, T206H | For Cessna U206F and TU206F: Engine: Centurion 4.0 BE-221 Propeller: MTV-9-D/210-58 Certification Basis: See CRI A-1 max. Take Off Power: 221 kW at 2300 min ⁻¹ (Propeller RPM) max. Continuous Power 221 kW at 2300 min ⁻¹ (Propeller RPM) Fuel: (...) | | 1. Supplement Pilot's Operating Handbook for the Cessna 208 equipped with TAE Centurion 4.0 installation; Doc. No.: 72-0310-72021, issue 1 or later approved issue; 2. Supplement Maintenance Manual Cessna 206 / Centurion 4.0, Doc. No.: AMM-72-01 (...) | | 19.08.2013 |
| 10015310 | THIELERT AIRCRAFT ENGINES GmbH | EASA.E.S.01000 | | | | Installation of the FADEC G6 in accordance with Installation Manual IM-80-01, Conversion Instruction CI-80-01 and Type Design Definition Doc. No.: TDD-80-01. | | Instructions for operation, maintenance and inspection: - Operation & Maintenance Manual Lycoming IO-580-B1A-E Converted Engine, Doc. No.: OM-80-01 - Installation Manual Lycoming IO-580-B1A-E Converted Engine, Doc. No.: IM-80-01 - Conversion Instruction for the Lycoming IO-580-B1A-E Engine | Engines having been converted via this STC may be identified as IO-580-B1A-E the data in the TCDS EASA.IM.E.027 remains valid, except the following: Weight of the engine equipped with FADEC: 207.49 kg (457.44 lbs) Dimensions: Overall Height: (...) | 03.08.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------------|-------------------|----------------|----------------------|---|--|---|---|---|---------------|
| 10036328 | THIELERT AIRCRAFT ENGINES GmbH | P-EASA.A.S.02566 | REV. 1 | | DA 40 D | Installation of engine TAE 125-02-114 in accordance with Conversion Instruction CI-50 02S DA 40 Issue/Revision: 1/0 and the Type Design Definition Document No: 50-01/Revision 1 follows a change of ownership. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: The scope of this STC complies with CS 23, initial issue dated 14 November 2003 plus Spec (...) | Flight Manual Supplement:- Anhang zum Flughandbuch Diamond DA 40 D mit TAE 125-02-114 Installation, TAE-Nr.: 50-0310-50011, Ausgabe 1-0. Supplement Pilot's Operating Handbook for Diamond DA 40 D equipped with TAE 125-02-114 installation, TAE-Nr.: 50-0 (...) | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product. | 19.08.2013 |
| 10042996 | THIELERT AIRCRAFT ENGINES GmbH | EASA.P.S.00580 | REV. 1 | | MTV-6-A | Installation of a Constant Speed Unit CSU TAE-125 Revision 1 follows a change of ownership. | | Manual for Installation and Operation Instruction - Constant Speed Unit CSU TAE-125, 02-6120-16001RX (X=Version No.), Issue 2, Revision 0 dated 27 April 2004 | 1. The CSU is limited to the case when the Propeller is driven by a TAE125 engine. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other ai (...) | 19.08.2013 |
| 10014379 | THOMSONFLY LTD. | EASA.A.S.01657 | | | BOEING 757-200 | Introduction of Security Cameras | | STC DEF 7571272AFM Supplement 752 Flight Vu. | | 07.12.2005 |
| 10014411 | THOMSONFLY LTD. | EASA.A.S.01706 | | | BOEING 767-300 | Introduction of BE Aerospace Spectrum Seat | | Thomsonfly Modification MOD/767/1462 Thomsonfly Modification Summary MS/767/1462 Issue 3 dated 27 February 2007 or later approved revision. | | 05.03.2007 |
| 10014461 | THOMSONFLY LTD. | EASA.A.S.02521 | | | BOEING 767-300 | Introduction of STG Aerospace Photo-Luminescent Passenger Emergency Escape Path Lighting System | | Thomsonfly Ltd MOD/767/1463 Issue 2 or later approved revision. STG Aerospace AFM Supplement M40-11 Revision A dated August 20 2002 or later approved revision. | Cabin & Exit Lighting levels must be measured for acceptability on each installation in accordance with Thomsonfly Technical Engineering Order TEO/767/1463 Issue 2 dated 26 April 2006 or later approved revision. | 02.05.2006 |
| 10042556 | THORNTON TECHNOLOGY Corp. | EASA.IM.A.S.00527 | | | DC-10-30F, MD-11F | Installation of a 9g cargo barrier net - ST00502LA. | | Drawing List 2550-1001, Rev. D, dated 7 July 2003, and 2550-1070.Rev.IR, dated 13 October 1997, or later approved Revision. | 1. As specified in Component Maintenance Manual 25-50-01, Rev. F, or later approved revisions. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility wi (...) | 18.01.2005 |
| 10043078 | THUNDER AVIATION SERVICES INC. | EASA.IM.A.S.03032 | | FAA STC ST01659LA | FAN JET FALCON FAN JET FALCON SERIES C, D, E MYSTERE-FALCON 20-C5, 2-D5 MYSTERE-FALCON 20-E5, F5 | EASA Validation of FAA STC ST01659LA Installation of Rosemount Pitot-Static Probes, Part Numbers 0856LR1 and 0856LR2 and Honeywell Air Data System comprised of dual AZ-252 Air Data Computers, dual BA-250 Altimeter Displays and AL-800 Altitude Altimeter (...) | ...FAA STC ST01659LA, as per FAA amendment 04th September 2007 *...Thunder Aviation N.A. Inc Master Drawing List 20-0100 Revision L dated 17th May 2007 *...For Aeroplanes equipped with Collins Model AP-105 Autopilot see "...EASA Approved Thunder Avii (...) | none | 18.05.2009 | |
| 10016435 | THUNDER BAY AVIATION LIMITED | EASA.IM.A.S.02775 | | TCCA STC SA04-55 | BOMBARDIER DHC-6 | Double and Single Utility Seats for DHC-6. | | Transport Canada approved Thunder Bay Aviation Master Drawing List, Doc. No. MDL013, Revision B, dated 27-Feb-2006, or later approved revision TAG Aerospace Maintenance Manual Supplement, Doc. No. 033002-07, Revision A, dated 08-Jun-2004, or later revision | | 19.03.2008 |
| 10043068 | THUNDER BAY AVIATION LIMITED | EASA.IM.A.S.03207 | | TCCA STC SA05-30 | DHC-6-100,-200,-300 SERIES | Installation, Substitute Composite Floor Panels. | | - Installation to be completed as per Aero Consulting Services Modification Data Summary E06100, Rev. E, January 2008, or later Transport Canada approved revision.- Instructions for continuing airworthiness as per Aero Consulting Services Report E06107, (...) | none | 06.05.2009 |
| 10033486 | TIMCO AEROSYSTEMS | | REV. 1 | FAA STC ST03143AT-D | 747-400, 747-400 (BCF), 747-400F | Installation of a Rockwell Collins Aero H+ Satcom System in accordance with TIMCO Master Data List 04T089L01, Rev. IR, dated November 25, 2005 or later FAA approved revision | The Certification Basis for the original product remains applicable to this certificate/ approval | N.A. | None | 04.02.2011 |
| 10035902 | TIMCO AEROSYSTEMS | | | FAA STC ST00552AT | B747-400 SERIES | Installation of a Collins Satcom update from 900 SDU to 906SDU (dataonly) in accordance with Page Avjet Master Drawing List 94-17, Initial Release, dated June 17, 1994, or later FAA approved Revision. , | The Certification Basis for the original product remains applicable to this certificate/ approval., | Master Drawing List Page Avjet 94-17, Initial Release, dated June 17, 1994 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | None. | 26.07.2011 |
| 10036655 | TIMCO AEROSYSTEMS | | | FAA STC ST01959AT-D | 747-400 | Installation of a Ball Camera Entertainment system per Timco Master Data List 99T057L001 Rev. C, dated November 8, 1999 or later FAA approved revision. Validation of FAA STC ST01959AT-D | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | none | none | 27.09.2011 |
| 10038130 | TIMCO AEROSYSTEMS | | REV. 1 | FAA STC ST03884AT-D | 767-300 | Installation of LCD Monitor Upgrade | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ Certification Basis as per the TCDS and CS 25 Amendment 5 for EWIS Requirements The requirements fo (...) | TIMCO MDL 10T024L001, Revision F, 24/02/2012 Instructions for Continued Airworthiness: 10T024R009, Revision C, 16/05/2011 10T024R509, Revision IR, 26/09/2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Deci (...) | None | 21.05.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------|-------------------|----------------|-------------------------|---|---|---|---|---|---------------|
| 10040972 | TIMCO AEROSYSTEMS | | | FAA STC NO. ST03928AT-D | 777-300 | Audio/Video On-Demand & In-seat Power System Installation. | The Certification Basis for the original product remains applicable to this certificate/ approval. | TIMCO Aerosystems Instructions for Continued Airworthiness 11T133R009, Rev. A dated September 29, 2011 or later revision of the above listed document approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | none | 10.08.2012 |
| 10044930 | TIMCO AEROSYSTEMS | | | FAA PROJECT ST14136AT-T | B767-300 | Floor Reinforcement in a Boeing 767-300 Series Aircraft | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Document 12T304R506 revision B Certification Plan - Floor Structure Reinforcement in a Boeing 767-300 series aircraft dated 11 March 2013; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (...) | STC only applicable for the following 767-300 Serial numbers: 26991, 26987, 26992, 26983, 26984, 26985, 26986, 26988 and 25209. The approval holder shall fulfill the obligations of Part 21, Paragraph 21A.109 or 21.A.451(b) (as applicable). Prior to instal (...) | 21.05.2013 |
| 10046684 | TIMCO AEROSYSTEMS | | | FAA ST04132AT | 767-300 | Floor Reinforcement in a Boeing 767-300 Series Aircraft | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS25 at Amendment 13:25.623 ; 25.625. The requirements for environmental (...) | TIMCO Aerosystems Master Data List: 12T304L001, Revision IR, 04/10/2013. TIMCO Aerosystems Instructions for Continued Airworthiness: 12T304R009, Revision IR, 09/08/2013 or later revisions of the above listed documents approved by EASA in accordance with (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 11.10.2013 |
| 10045384 | TIMCO AEROSYSTEMS, L.L.C. | | | FAA STC ST01735AT-D | MD-11 | VHF Installation - MD-11 Collins VHF Transceivers and Cables/Radio Tuning Panels installation to enable 8.33 kHz channel spacing in an MD-11 aircraft | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | TIMCO Master Data List No. 98T058L001 Rev. B dated 29 October 1998 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) # and/or the Technical Implem (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product (...) | 20.06.2013 |
| 10015495 | TIMCO AVIATION SERVICES INC. | EASA.IM.A.S.01257 | REV. 1 | FAA STC ST03068AT-D | BOEING 767-300 | Cabin Standardization in accordance with TIMCO Master Data List 03T012L001, revision D, dated November 30th, 2006, or later FAA approved revision. | | | This approval should not be extended to other aircraft of this model on which other previously approved modifications are incorporated unless it is determined by the installer that the interrelationship between this change and any of those other prev (...) | 01.12.2006 |
| 10015545 | TIMCO AVIATION SERVICES INC. | EASA.IM.A.S.01433 | | FAA STC ST03197AT | BOEING 737-800 | Complete interior reconfiguration in accordance with TIMCO Aviation Services, Inc. MDL report 04T101L001, revision "C", dated December 07, 2005, or later EASA approved revision | | - MDL report 04T101L001, revision "C", dated December 07, 2005- TIMCO report 04T101R007, Instructions for Continued Airworthiness - Interior reconfiguration, Revision "IR", dated February 9, 2005, or later EASA approved revisions. | 1. Compatibility of this design change with other previously approved modifications must be determined by the installer. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragr (...) | 17.07.2006 |
| 10015601 | TIMCO AVIATION SERVICES INC. | EASA.IM.A.S.01537 | | FAA STC ST03107AT-D | BOEING 767-300 | Interior Reconfiguration | | TIMCO Master Data List 04T110L001 Revision IR dated December 08 2005 or later approved revision | | 15.12.2005 |
| 10015636 | TIMCO AVIATION SERVICES INC. | EASA.IM.A.S.01592 | | FAA STC ST01929AT-D | BOMBARDIER DHC-8-100 SERIES BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-300 SERIES | Installation of Flight Data Recorder Expanded Parameters with a Plessey FDAU | | Master Data List 99T037L001 Rev. L or later approved revision | Limited to: 99T037C002 for DHC-8-100/200/300 and 029C009 for DHC-8-300. Maintenance required: pitch, roll and yaw position transducers and strings have to be detailed visually inspected and checked for smooth operation acc. MRB interval 2C | 19.07.2006 |
| 10015637 | TIMCO AVIATION SERVICES INC. | EASA.IM.A.S.01593 | | FAA STC ST03135AT-D | BOEING 767-300 | Installation of Rockwell Collins ACARS CMU-900 | | TIMCO Master Data List 05T036L001 Revision IR dated November 21, 2005 or later approved revision | | 28.11.2005 |
| 10015831 | TIMCO AVIATION SERVICES INC. | EASA.IM.A.S.01922 | REV 1 | FAA STC ST03372AT | BOEING 767-300 | Installation of Lower Lobe Crew Rest Module (CRM) and Main Deck/Vestibule in accordance with Timco Master data List 04T104L001, Rev H, dated December 7, 2007 or later approved revisions approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). | | 1. Approved Airplane Flight Manual Supplement TIM-AFM-01044 Rev. A, dated December 14, 2007. 2. Instructions for continued Airworthiness 04T104R007, Rev A, dated November 9, 2006. | 1. This approval should not be extended to other Aircraft of this model on which other previously approved modifications are incorporated unless it is determined by the installer that the interrelationship between this change and any other previo (...) | 16.01.2008 |
| 10015910 | TIMCO AVIATION SERVICES INC. | EASA.IM.A.S.02037 | | FAA STC ST03107AT-D | BOEING 767-300 | Interior Reconfiguration | | - TIMCO Master Data List 04T110L001 Revision C dated May 31 2005 or later approved revisions- TIMCO Instructions for Continued Airworthiness 04T110R115 Revision IR dated May 12 2006 | | 16.08.2006 |
| 10015974 | TIMCO AVIATION SERVICES INC. | EASA.IM.A.S.02125 | | FAA STC ST03300AT-D | AIRBUS A321 | Installation of a doghouse (storage container) in the bulk cargo compartment. | | Master Data List, Document Number 05T101L001, Revision a, dated 26 September 2006, or later EASA approved revision. Instructions for Continued Airworthiness, Document Number 05T101R006, Revision IR, dated 06 September 2006, or later EASA approved revision. | Limitations and Conditions specified in FAA STC ST03300AT-D remain applicable. | 18.01.2007 |
| 10015998 | TIMCO AVIATION SERVICES INC. | EASA.IM.A.S.02155 | | FAA STC ST03342AT-D | BOEING DC-9-87 (MD-87) BOEING MD-88 | Installation of a Door Safety Strap | | TIMCO Master data list 06T 005L001, Revision B dated October 11, 2006 or later FAA approved revision | 1. TIMCO's Instructions For Continued Airworthiness 06T005R014, Rev. IR dated June 8, 2006, is a required part of this STC. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the par (...) | 24.10.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|-----------------------------------|-------------------|----------------|-------------------------|---|---|--|---|---|---------------|
| 10016002 | TIMCO AVIATION SERVICES INC. | EASA.IM.A.S.02163 | | FAA STC ST01216AT-D | AIRBUS A300 | Installation of Airshow 210 System | | TIMCO Master data list, MDL-01094, Rev. L Configuration Document, CFG-01105, Rev. E | -- | 08.10.2006 |
| 10027196 | TIMCO AVIATION SERVICES INC. | | | FAA STC ST00696AT-D | B747-400 | Installation of Collins SAT-906 SatCom System with Ball Antenna in accordance with Timco Master Data List MDL-01068, Revision M, dated October 27, 1999 or later approved revision. | The Certification Basis for the original product remains applicable to this certificate/ approval | None | None | 14.09.2009 |
| 10030407 | TIMCO AVIATION SERVICES INC. | | | FAA STC ST03820AT-D | B767-300 | Interior reconfiguration, B767-300 validation of FAA STC ST03820AT-D The aircraft will have the seating configuration changed to additional coach (Y) class seating, additional seat track installation with associated structural modifications, removal (...) | As per EASA TCDS IM.A.035 For this STC the following CRI/Special Condition is applicable and is part of the certification basis: CRI H-01 "Enhanced Airworthiness Programme for Aeroplane System - ICA on EWIS" The requirements for environmental protection a (...) | -Master Data List 09T116L001, Rev IR, dated June 11th, 2010-TIMCO's Instructions for Continued Airworthiness 09T116R009, Rev IR dated April 23, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2 (...) | As per FAA STC ST03820AT-D | 15.06.2010 |
| 10014179 | TMC AERO | EASA.A.S.01316 | | | AIRBUS A330-243 | This STC deals with the installation of a TMC stretcher. All documents related to this STC are listed in the STC form A330-25-23. The associated "Fiche de Modification" from Corsair is referenced 25-137 Rev. 1 (or later EASA-approved revision) | | | The approval of this modification is limited to the Airbus A330-243. the limitations and conditions defined in the EASA Datasheet are applicable. | 02.06.2005 |
| 10014180 | TMC AERO | EASA.A.S.01317 | | | BOEING 737-400 | Temporary Installation of a Stretcher that is fitted on the Passengers Seat Track | | -TMC Aero Airplane flight Manual Supplement Doc Ref AFMS25-0317, Rev 1 dated 11th August 2005-TMC Aero "Fiche de Modification", Doc Ref 25-135, Rev 5 dated 11th August | 1. Operation of the aircraft with stretcher installed is limited to 100 FH per year.2. The limitations as described in the AFM Supplement Doc Ref AFMS25-0317 apply | 29.08.2005 |
| 10014181 | TMC AERO | EASA.A.S.01318 | | | BOEING 747 | The installation of stretchers mounted on the seat tracks in passenger cabin of the Boeing 747 for the transport of patients, as described in Corsair Fiche de Modification no. 25-136 | | - Corsair Note Technique DT/ET/044102/NT-Corsair Fiche de Modification no. 25-136-JTSO F.O. 103 for model TMC 99 | 1. The maximum number of passengers allowed when two stretchers are installed shall be determined.2. When three or more stretchers are installed transport of passengers is not allowed. | 12.07.2005 |
| 10033808 | TOST GMBH FLUGZEUGGERÄTEBAU | EASA.A.S.03948 | | LBA STC 0326/XXX | HUSKY A-1A, A1-B | Installation of Tow Cable retractor Winch with Guillotine to Husky A-1A,A-1B. Extension of LBA STC 0326/xxx, dated 21 October 1992. This STC is a revision to and supersedes LBA STC0326/xxx, dated 21 October 1992. | CS-23.C | -Anhang zum Flughandbuch, LBA approved 24 January 1985 is now valid for Husky A-1A and A-1B: Flug- und Betriebsanweisung für Flugzeugabschleppsel-Einziehwinde mit Kappvorrichtung für Piper, Husky, Citabria, Maule, Stinson-The following Installation Man (...) | None | 23.06.2009 |
| 10028946 | TOAS - TOTAL QUALITY AND AIRCRAFT | | | | B737-500 | Installation of Floor proximity emergency photo luminescent path. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval Additional Certification Requirements as agreed for FAA and Transport Canada have been a (...) | Design Dossier CG-008-09-CPO dated February 11th 2010.AFM Supplement CG-008-09-AFM Iss.1 or later revisions of the above listed documents approved by EASA | None | 22.02.2010 |
| 10031790 | TOAS - TOTAL QUALITY AND AIRCRAFT | | | | A310-304 | Cabin Modification to Full Economy Cabin Reconfiguration from a 18C/184Y seats configuration to a 224Y seats configuration | The Certification Basis for the original product remains applicable to this certificate/ approval The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Design Dossier (->-) ref. CG-001-010, dated 08 June 2010, Change/Repair Approval Sheet, ref. CG-001-10-MAS Issue 3, dated 07 June 2010, Service Bulletin, ref. CG-001-10-SRB Issue 5, dated 07 June 2010, or later revisions of the above listed documents approved by EASA | Limited to MSN 573 | 14.09.2010 |
| 10014421 | TRANS AIRCO S.A. | EASA.A.S.01742 | | FOCA STC 23-15-001 ISS2 | PILATUS PC-12 PILATUS PC-12/45 PILATUS PC-12/47 | Aircell 3100 Iridium SATCOM Installation | | | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 07.09.2006 |
| 10014671 | TRANS AIRCO S.A. | EASA.A.S.02831 | | | PILATUS PC-12 PILATUS PC-12/45 PILATUS PC-12/47 | F.D.S.FD932 DVD Player Installation | | Master Certification Documents List TSA PC12-23.33.02 MCL | None | 08.11.2006 |
| 10014751 | TRANS AIRCO S.A. | EASA.A.S.02936 | | | MYSTERE FALCON 900 | MODE S TRANSPONDER WITH ENHANCED SURVEILLANCE CAPACITY | | Master Certification Document List TSA F900-34.80.03 MCDL, Rev. Original or later approved revisions Airplane Flight Manual Supplement TSA F900-34.80.03 AFMS, Rev. Original or later approved revisions Maintenance Manual Supplement TSA F900-34.80.03 MMS, Rev. Original or later approved revisions | This STC is approved only for those Mystere Falcon MF-900 aircraft excluding Falcon 900C version and incorporating at least: # Dual Collins TDR-94D Mode 4S# ATC Transponders compliant with Elementary Surveillance and# Single or dual GPS receive (...) | 13.02.2007 |
| 10014789 | TRANS AIRCO S.A. | EASA.A.S.02990 | | FOCA STC 31-31-05 | FAN JET FAN JET FALCON SERIES C,D,E,F MYSTERE FALCON 20 C5, D5, E5 MYSTERE FALCON 20 F5 | AVIONICA mini QARA VIONICA mini QUAR P/N Note: This STC supersedes FOCA STC FOCA 31-31-05, dated 09.09.2002 | | TSA Engineering Order TSA F20-31.62.01EO, Rev. A | 1. Valid for Dassault Aviation:- Fan Jet Falcon Basic- Fan Jet Falcon Series C, D, E, F, Mystere-Falcon 20 Series C5, D5, E5, F52. This STC is approved only for the product configuration as defined in the approved design data referred to in the para (...) | 23.01.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------------------|---------------------|----------------|------------------------------|---|--|--|--|---|---------------|
| 10015474 | TRANS AIRCO S.A. | EASA.IM.A.S.01226 | | | MYSTERE FALCON 50 | Replacement of existing HONEYWELL MK II GPWS and COLLINS MFD-85 with a UNIVERSAL Terrain Awareness and Warning system (TAW/S) and a UNIVERSALMFD-640 | | Swiss FOCA AFMS UNIVERSAL Terrain Awareness and Warning System and MFD-640 Multi-Function Display TSA F50-31.80.05AFMS dated 23.03.2005 or later approved version.Swiss FOCA Instructions for Continued Airworthiness (ICA) TSA F50-31.80.05MMSdated 23.03.2005 or later approved version. | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer.2. Limited to Falcon 50, S/N 159, HB-1SD | 19.08.2005 |
| 10015621 | TRANS AIRCO S.A. | EASA.IM.A.S.01573 | | | FALCON 20 | Incorporation of Mode S EHS capability | | Modification TSA F20-34.80.01AFMS F20-34.80.01 AFMS | 1. Limited to Falcon 20 S/N 237; 257; 307; 341; 389; 4262. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 14.09.2006 |
| 10015700 | TRANS AIRCO S.A. | EASA.IM.A.S.01688 | | | PC-12, PC-12/45 | Structural installation of Honeywell KA-815 TCAS I Antennas | | Engineering order TSA PC12-34.84.02EO | | 09.03.2006 |
| 10015753 | TRIAD INTERNATIONAL MAINTENANCE | EASA.IM.A.S.01786 | | FAA STC ST03200AT | BOEING 737-800 | Complete interior reconfiguration in accordance with TIMCO Aviation Services, Inc MDL, report 04T101L500, revision #D#, dated December 08, 2005, or later EASA approved revision | | - MDL, report 04T101L500, revision "D", dated December 08, 2005 - TIMCO report 04T101R014, Instructions for Continued Airworthiness - Interior reconfiguration, Revision "R", dated March 01, 2005, or later EASA approved revisions. | | 20.07.2006 |
| 10016092 | TRI-STAR AVIATION INC. | EASA.IM.A.S.02279 | | FAA STC ST01897CH & EASA 200 | L1011-385-3 | Installation of strengthened flight deck doorNote: This STCsupersedes EASA STC 2004-3731. | | Tri-Star Aviation Master Data List MDL-501008 Rev. J dated 10/01/2005 | 1. Limited to S/N 193H-1206, 193H-1248, 293B-1240, 239B-1241, 193W-1189, 193Y-1181, 193Y-1184, 193Y-1185, 193Y-1176,193Y-1194, 193Y-1208, 193H-1202, 193H-1207, 193H-1216 and 193H-12182. This STC is approved only for the product configuration a (...) | 08.01.2007 |
| 10029274 | TRIUMPH INSTRUMENTS BURBANK Inc. | EASA.IM.A.S.03255 | REV. 1 | FAA STC ST02236LA | CL-600-2B19, CL-600-2C10 CL-600-2D15, CL-600-2D24 | Triumph Replacement Hydraulic Accumulators. | EASA.IM.A.023.The Certification Basis for the original product remains applicable to this certificate/ approval.The certificated noise and/ or emissions levels of the original productare unchanged and remain applicable to this certificate/ approval. | EASA STC No. EASA.IM.A.S.03255 dated 27 July 2009.FAA STC No. ST02236LA dated 23 April 2009.FAA Approved Model List (AML) ST02236LA dated 29 December 2009.Instructions for Continued Airworthiness (ICA), Triumph Master ReportNumber ICAW-MR-ST13664, R (...) | None. | 17.03.2010 |
| 10034472 | TRIUMPH INSULATION SYSTEMS, L.L.C. | | REV. 1 | FAA STC ST02013LA | B767-200 | Insulation Blanket Installation | The Certification Basis for the original product remains applicable to this certificate.The requirements for environmental protection and the associatedcertificated noise and/ or emissions levels of the original product areunchanged and remain applicable to this approval. | Meximl Company#s Index List, Report Number B2000-3 RevisionA, 22/07/2008or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 11.04.2011 |
| 10026913 | TROLLTUNE CORPORATION | P-EASA.IM.A.S.03179 | | FAA STC SA03608AT | 182P AND 182Q (SEE ATTACHMENT) | Increased Maximum Gross Takeoff Weight (MGTOw). This STC allows a 150 lbincrease in MGTOw from 2950 to 3100 lbs. There is no change to themaximum landing weight of 2950 lbs. | CAR 3 as defined on FAA STC and CS-36. | In accordance with Trolltune Corporation Master Data List (MDL), Document Number EDS000-SW, Revision 6, dated August 20, 2008 or later approved revisions.Operation in accordance with Trolltune Corporation Airplane FlightManual Supplements (AFMS), dated (...) | None. | 21.08.2009 |
| 10013953 | TROYES AVIATION | EASA.A.S.00493 | REV. 2 | | CESSNA F177-RG | Installation of sliding panel on the front left and right door windows and rear right window.Addition of new S/N 087 F-BURU on STC EASA.A.S.00493. | | Minor Change No. 52.Major Change No. 42 approved under EASASTC EASA.A.S.00493 Revision 1.Flight Manual Supplement No. 42, Revision 2, 5 January 2008.Compliance Statement dated 5January 2008. | Limited to F177RG S/N 0112, 0065 and S/N 087 F-BURU. | 28.02.2008 |
| 10013954 | TROYES AVIATION | EASA.A.S.00494 | REV. 3 | | CESSNA F177RG | Installation of a camera sight door under the fuselage.Addition of new aircraft to STC EASA.A.S.00494 Revision 2. | CS23The Certification Basis for the original product remains applicable to this certificate/ approval. | File ref: FAM N° 59CHANGE FILE N° 53 ISSUE JUNE 2006 - EASA.A.S.00494 R1 and R2MINOR CHANGE N°41 | Applicability F177RG S/N 0112, 0065, 103, 087 F-BURU and S/N 51 F-HBAG. | 29.03.2010 |
| 10014266 | TROYES AVIATION | EASA.A.S.01495 | REV. 3 | | REIMS CESSNA F177-RG | Provisions for three attachments in the right side of the instrumentpanel in accordance with the STC EASA.A.S.01495R1: adjunction of a newserial number "S/N 051" | CS-23 (FAR 23) | -TroYES Aviation REF 51.-Minor Change TroYES Aviation REF 51 Revision 2 July 2009-Minor Change TroYES Aviation REF 45 approved under EASA.A.S.01495Revision 1.- Supplement Flight Manual Rev. 1, dated 18/11/2005. | Limited to: F177RG S/N 0112, 0065, 103, 153 and 087 F-BURU;S/N 051D-EMEN | 06.10.2009 |
| 10014285 | TROYES AVIATION | EASA.A.S.01522 | | | CESSNA 340A | FAM n° 47 - Limitation of MTOW to 2699 kg | | Flight Manual Supplement "A" Edition 1 Revision 0 dated December 05. | s/n 340A-0451 | 23.02.2006 |
| 10014642 | TSA TRANS AIRCO S.A. | EASA.A.S.02792 | | | PILATUS PC-12 PILATUS PC-12/45 PILATUS PC-12/47 | AFT Refreshment Galley and OvenRefer to master certification documents list TS PC12-25.30.02MCL | | Airplane Flight Manual Supplement TSA PC12-25.30.02AFMS | 1. Limited to serial number 321 and later;2. Galley power and microwave oven use is not approved during take-off and landing phases of flight and during emergency or abnormal procedures;3. When aft refreshment galley is installed,the luggag (...) | 21.11.2006 |
| 10015718 | TTF AEROSPACE, LLC | EASA.IM.A.S.01716 | | FAA STC ST00231BO | BOEING 737-800 | Installation of Yokohama forward Lavatory E and Yokohama aft lavatory K and attendant seat in accordance with TTF Aerospace MDL TTF-05T026-01 Rev B, dated February 24, 2006 | | -TTF Aerospace MDL TTF-05T026-01 Rev B, dated February 24, 2006- Instruction for continued airworthiness (ICA), TTF Aerospace Document TTF-05T026-18 revision new, dated February 13, 2006 or later FAA accepted revision | Limited to installation on Boeing 737-800 S/N 28325, 30470,32277 and 32278 only.Installation of FAA STC 01533LA requiredInstruction for airworthiness 9ICA, TTF Aerospace DocumentTTF-05T026-18 revision new, dated February 13, 2006 or later FAA accept (...) | 06.04.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|-------------------|----------------|-------------------------------|---|---|--|---|--|---------------|
| 10016453 | TTF AEROSPACE, LLC | EASA.IM.A.S.02799 | | FAA STC ST01195SE | BOMBARDIER DHC-8-100 SERIES BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-300 SERIES | Installation of a Reinforced Flight Deck Door. | | TTF Aerospace, LLC Master Data List (MDL), TTF-MDL-02T011, Rev. J, dated 05.10.2007.Or later approved revisions of the above document approved by EASA in accordance with EASA ED Decision 2004/04/CF or subsequent revisions of this decision. | | 16.06.2008 |
| 10045055 | TUERK HAVA YOLLARI TEKNİK A.S. | | | | A321-231 | CONVERSION OF CABIN LAYOUT FROM 210 PAX FULL E/C TO 194 PAX DUAL CLASS AND MOVABLE CLASS DIVIDER (MCD) INSTALLATION. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Design Declaration CG-2520005-DD01-01, dated 20.04.2013 or later revisions of the above listed documents approved by EASA | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 27.05.2013 |
| 10046655 | TUERK HAVA YOLLARI TEKNİK A.S. | | | | A340-311 | CABIN LAYOUT CONVERSION FROM DUAL CLASS 270 PAX TO SINGLE CLASS 331 PAX | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Design Declaration Form C9-2520006-DD01-02, Dated 10-07-2013 or later revisions of the above listed documents approved by EASA | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 08.10.2013 |
| 10015250 | TUIFLY GmbH | EASA.A.S.03692 | | | B737-800 | Cabin Configuration Change to 52BC plus 6YC | | Master Data List B7378-25-08-02 Rev.06 Cabin Layout Drawing 25-378105 R2 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. | 05.11.2008 |
| 10015258 | TUIFLY GmbH | EASA.A.S.03711 | | | B737-700, B737-800 | Installation of Water Boiler P/N: 66601-101-0001 as an alternative to Coffee Maker P/N: 64753-001-003. | | - Hapag-Lloyd Master Data List MDL B737NG-25-08-07 Rev.2 dated 19.01.2009 or later approved revisions. - Component Maintenance Manual CMM 25-37-13. | 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product (...) | 15.04.2009 |
| 10017090 | TURBOMECA | EASA.R.S.00753 | REV. 2 | | AS350B, B1, BA, B2, B3 EC130B4 | Installation of Sand Filter (Engine air inlet protection) Notes: Revision 1 to this STC reflects: 1. the transfer of EASA STC EASA.R.S.00753 from CGTM to Turbomeca; 2. updates of the Associated Technical Documentation package following DOA minor changes R (...) | As per EASA TCDS.R.008 Issue 04, 23 November 2009. The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Associated technical documentation, including definition and installation documentation, maintenance documentation, and operating documentation, as per the attached continuation sheet. Note: RFMSs at revisions as specified in the attached continuation sheet or later revisions approved by EASA | None. | 17.06.2010 |
| 10017287 | TURBOMECA | EASA.R.S.01536 | | DGAC FRANCE STC NO 84-SF-1122 | EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 BA | Installation of HESIS thermic camera system manufactured by SAGEM | | - Definition: CGTM technical note NT 0339/PR/c # 25/02, dated 20.06.2002 - Installation: CGTM technical instruction IT 0339/JYC/c # 43/01, dated 21.05.2002 and SAGEM Note, HESIS-GN5100/Doc/007/SAG.- Maintenance: CGTM technical note NT (...) | 1. Only Turbomeca personal or specifically trained mechanics are allowed to install fixed parts and components of the HESIS kit. 2. Prior to installation of this modification the installer must determine that the interrelationship between this mod (...) | 05.01.2009 |
| 10016861 | TYLER CAMERA SYSTEMS | EASA.IM.R.S.01237 | REV. 1 | FAA STC SR01272LA | R44, R44 II | Installation of Tyler ROB-001 Universal Mount | | 1. Flight Manual Supplement ROB-RFM-001 Rev B, dated 28/11/2001 or later FAA approved revision. 2. Installation Manual R44INST-001, Rev. B dated 28/11/2007 or later FAA approved Revision. 3. Tyler Camera Systems Master Drawing List No. TCS R-44-001, Rev (...) | | 04.12.2007 |
| 10016594 | U.S. TECHNICAL CONSULTANTS, INC. | EASA.IM.A.S.03005 | | FAA STC ST02409CH-D | DC-9-83 (MD-83) DC-9-87 (MD-87) | FAA STC No. ST02409CH-D #NavAero t-BagC22 Electronic Flight Bag# | | MDL 4696-23-000 Rev. F dated 13 August 2007 FM Supplement U10-0163 dated 6 November 2007 | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The (...) | 16.01.2009 |
| 10029398 | U.S. TECHNICAL CONSULTANTS, INC. | | | FAA STC ST02161LA | B757-200, B757-300 | Installation of navAero t.BagC2 Electronic Flight Bag | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -FAA STC ST02161LA-Master Drawing List 4913-23-000 rev B dated 05/02/08-AFMS U-10-0171 dated 08/11/08 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | Not Applicable | 23.03.2010 |
| 10033897 | U.S. TECHNICAL CONSULTANTS, INC. | | | FAA STC ST02320LA | B767-200, -300, -300F, -400ER | Installation of Class 2 Electronic Flight Bag (EFB) provisions. | The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this approval. | US Technical Master Drawing List (MDL), Document No. 4914-23-000, Revision C, 20/08/2010 Aeroplane AFM Supplement, Document No. U10-0175, Revision NC, 27/10/2010 Instructions for Continued Airworthiness, Document No.U10-0195, Revision NC, 08/04/2010 or I (...) | Any equipment that is connected to the electrical power provisions certified by this STC must be tested in accordance with the requirements of navAero Inc.# Electrical and Environmental Substitution for Lithium based battery in navAero t-bag C22 EFB (...) | 17.02.2011 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|----------------|----------------|----------------------|------------------------------|---|---------------------|---|--|---------------|
| 10014236 | UNI AIR ENTREPRISE | EASA.A.S.01399 | | | CORVETTE SN 601 SN 028 & 008 | Deviation to SB 34-12 replacement of mount C24-C25 in accordance with MDL UAE 008.05 Rev. 1 dated 8/07/2005. Installation of horizontal mount between frames C24 and C26 instead of vertical one between frames C24 and C25. | | AFMS No. 58 | 1. This STC applies to MSN 028 and MSN 008 only.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined. | 25.07.2005 |
| 10013932 | UNI AIR ENTREPRISE SAS | EASA.A.S.00305 | REV. 1 | | CITATION 550 | - Installation of Digital Data Recorder SSUFDR Allied Signal (TSO-C124) in accordance with Master Data List No. UAE.011.02 Issue0 dated 16 August 2004, or later EASA approved Revision.- Installation of a Digital Flight Data Recorder.Note: Revision 1 of STC EASA.A.S.00305. | | UAE 011 02 ED 3 Rev 7 | 1. AFM Supplement UAE No. 46 Revision 0 dated 16 August 2004 or later EASA approved revision is a required part of this STC.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the par (...) | 06.02.2009 |
| 10014004 | UNI AIR ENTREPRISE SAS | EASA.A.S.01041 | | | BEECH C90A | Installation of a Concorde RG-380E Sealed Lead Acid Battery. | | -FAA STC n° SA00257AT.-United Beechcraft Master Drawing List MDL-UBF93101 (15/06/1993). FAA Form 337 No ML2R009L. | 1. Supplemental Airplane Flight Manual No UAE/52 (French language) must be used with this modification.2. Modification limited to Beechcraft C90A s/n LJ-1361.3. This STC is approved only for the product configuration as defined in the approved design (...) | 11.04.2005 |
| 10014805 | UNI AIR ENTREPRISE SAS | EASA.A.S.03008 | | | FALCON 20 | CABIN INTERIOR REFURBISHMENT | | Dossier de Demonstration de Conformite DDC-002-07 Revision 1 or later approved revisions.Master Data List MDL.002.07 Revision 1 or later approved revisions.Aircraft Flight Manual Supplement UAE-AFMS-07-001 Original Issue or later approved revisions. | This STC is approved only for Mystere Falcon 20-E5 aircraftserial number 302.Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification (...) | 15.05.2007 |
| 10014824 | UNI AIR ENTREPRISE SAS | EASA.A.S.03037 | | | BEECH 300 BEECH 300C | FMS upgrade from UNS-1K to UNS-1K+ for compliance with P-RNAV | | Master Data List, Doc. No. MDL.006.07, Revision 0, dated 03-Mar-2007, or later approved revisionEASA approved Airplane Flight Manual Supplement for Beech 300, Doc. No. UAE-AFMS-07-003, Revision 0, dated 08-Mar-2007, or later approved revision | This change is limited to the following serial numbers:B300: S/N FL-111 thru FL-380; and FL-382B300C: S/N FM-9thru FM-11; and FN-2 and subsequent.This change is limited to the configuration of aircraft / equipment type as listed under Item 1 (...) | 30.04.2007 |
| 10014825 | UNI AIR ENTREPRISE SAS | EASA.A.S.03038 | REV.1 | | BEECH 300 BEECH 300C | Dual Installation of Collins TDR-94D with CTL-92E for compliance with Enhanced Surveillance capability. | | Master Data List, Doc. No. MDL.007.07, Revision 3, dated 07-September-2007, or later approved revision. | This change is limited to the following serial numbers:B300: S/N FL-111 thru FL-380; and FL-382B300C: S/N FM-9 thru FM-11; and FN 2 and subsequent.This change is limited to the configuration of aircraft/equipment type as listed under Item 1 - Descri (...) | 11.09.2007 |
| 10014853 | UNI AIR ENTREPRISE SAS | EASA.A.S.03077 | | | BEECH B300 | Installation of a Digital Flight Data Recorder | | Master Data List ref. MDL-012-07 revision 1, dated April 20, 2007 or later approved revisionAirplane Flight Manual Supplement ref. UAE-AFMS-07-003 revision IR, dated March 8, 2007or later approved revision | 1. This STC is limited to the following S/N:Beech B300 S/N FL-111 through FL-380, FL-382Beech B300C S/N FM-9 throughFM-11, FN-2 and after2. Prior to installation of this modification the installer must determine that the interrelationship betw (...) | 06.06.2007 |
| 10014875 | UNI AIR ENTREPRISE SAS | EASA.A.S.03100 | | | | Upgrade of ATC transponders for elementary surveillance (Mode S). To add elementary surveillance functionality in the ATC transponder and replace KFS-578A control box by PS-578 to comply the ECAC airspace requirements in accordance with Master Data List MDL-010-07, dated 11 May, 2007. | | Data package containing the following documents, and additional supporting data:MDL-010-07 Master Data List Modification UAE-010-07.DDC-010-07 Compliance Design Document Modification UAE-010-07.UAE-CD-E-0008-07 Ground Test Functional Procedure for a (...) | | 06.09.2007 |
| 10014951 | UNI AIR ENTREPRISE SAS | EASA.A.S.03215 | | | MYSTERE FALCON 50 | Antenna and Cables installation provisions for satellite communication system. | | 1. Uni Air Enterprise Master Data List MDL-016-07 Revision 4, dated 10 June 2008 .2. Installation instructions in accordance with Unir Air Enterprise Service Bulletin SB-UAE-016-07-01 Revision 3, dated 9 June 2008 ;3. Unir Air Enterprise Mai (...) | 1. STC nor applicable to Falcon 50EX version aircraft.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce (...) | 18.06.2008 |
| 10015156 | UNI AIR ENTREPRISE SAS | EASA.A.S.03523 | | | MYSTERE FALCON 20 | Installation of a forward partition door between entry areaand cabin | | Modification i.a.w Uni Air Enterprise Master Data List MDL-010-08 Revision 1, dated 14th May 08 or later approved revisions.Uni Air Enterprise Airplane Flight Manual Supplement UAE-AFMS-08-001 Edition 01, dated 13th May 2008 or later approved revisions. | Design change applicable to Mystere Falcon 20 aircraft serial number 311. | 19.05.2008 |
| 10015172 | UNI AIR ENTREPRISE SAS | EASA.A.S.03546 | | | MYSTERE FALCON 20 | Installation of a passenger entertainment audio and video system in the cabin. | | Uni Air Enterprise Master Data List MDL-008-08 Revision 1, dated 15th May 2008.Installation instructions in accordance with Uni Air Enterprise Service Bulletin SB-UAE-008-08-001 Revision 1, dated 16th May 2008.Uni Air Enterprise Airplane Flight Manu (...) | Design change applicable to Mystere Falcon 20 aircraft serial number 311. | 19.05.2008 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|------------------|----------------|----------------------|--------------------------|---|--|---|--|---------------|
| 10015241 | UNI AIR ENTREPRISE SAS | P-EASA.A.S.03669 | REV. 1 | | MYSTERE FALCON 50 (MF50) | Iridium telecommunication system installation. This change is a revision of STC EASA.A.S.03669. This change allows installation of an alternative P/N of transceiver and handsets as well as alternative location. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Uni Air Enterprise Master Data List MDL-016-08 Revision 2, dated 20th July 2010. Installation instructions in accordance with Uni Air Enterprise Service Bulletin SB-UAE-016-08-01 Revision 0, dated 26th September 2008. Uni Air Enterprise Airplane Flight (...) | Installation limited to those Mystere Falcon 50 (MF50) aircraft previously modified i.a.w. Uni Air Enterprise S.A.S. Supplemental Type Certificate (STC) "Antenna and cables installation provisions for satellite communication system" approved under EASA (...) | 13.08.2010 |
| 10031744 | UNI AIR ENTREPRISE SAS | | | | MYSTERE FALCON 20 | Enhanced Mode "S" compliance upgrade. Transponder and FMS software upgrade to meet the requirements for mode "S" Enhanced Surveillance downlink parameters. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Uni Air Enterprise Master Data List MDL-011-10 Revision IR, dated 26th August 2010. Uni Air Enterprise Compliance Design Document DDC-011-10 Revision IR, dated 23rd August 2010. Uni Air Enterprise Airplane Flight Manual (AFM) Supplement UAFMS-10-002 Rev (...) | None | 09.09.2010 |
| 10031773 | UNI AIR ENTREPRISE SAS | EASA.A.S.00307 | REV. 1 | | CESSNA 550 (CITATION II) | Installation of Honeywell EGPWS Mark VIII. The STC has been revised to implement the alternative EGPWS computer P/N 965-1210-020 and the alternative EGPWS Configuration Module P/N 700-1710-020. | EASA TCDS IM.A.207. The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are (...) | Master Data List MDL-004-02, Revision 6 from 08 September 2010. Aeroplane Flight Manual Supplements-AFMS UAE No. 15, Edition 2, Revision 2 from 24 May 2004 for S/N 550-0609-AFMS UAE No. 15, Edition 2, Revision 6 from 06 April 2004 for S/N 550-0190-AFMS U (...) | Limited to: S/N 550-0001 through 550-0733. The maintenance tasks referred in Supplement to Cessna 550 Maintenance Manual ref. EGPWS UAE No 4 Revision 6 are required. The audio de-clutter function must be disabled. | 10.09.2010 |
| 10037035 | UNI AIR ENTREPRISE SAS | | | | 200 | Installation of gear override and steep approach switches on TWASTW-950. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement No. UAE-AFMS-11-002 revision IR dated 18/07/2011. Master Data List No. MDL-009-11 revision 1 dated 13/09/2011. Service Bulletin No. SB-UAE-009-11-01 dated 18/07/2011 or later revisions of the above listed documents approved by EASA. | This STC is applicable to aircraft s/n BY-91. | 26.10.2011 |
| 10038465 | UNI AIR ENTREPRISE SAS | | | | MF50 | Enhanced Mode "S" compliance upgrade. This modification consists in the upgrade from mode "S" transponder elementary surveillance to mode "S" transponder enhanced surveillance to meet the European airspace mandate for mode "S" EHS downlink parameters. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Uni Air Enterprise Master Data List MDL-002-11 Revision 2, dated 09th February 2012. Uni Air Enterprise Airplane Flight Manual Supplement (AFMS) UAE-AFMS-11-001 Revision IR, dated 09th December 2011, or later revisions of the above listed documents approved by EASA. | None | 17.02.2012 |
| 10039277 | UNI AIR ENTREPRISE SAS | | | | MF 20GF | Installation of Patch Iridium Antenna. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Uni Air Enterprise Master Data List MDL-011-11 Revision 5, dated 30th March 2012, or later revisions of the above listed documents approved by EASA. | None | 23.04.2012 |
| 10041337 | UNI AIR ENTREPRISE SAS | | | | MF900 | Installation of Iridium Satcom System ICS-120B. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Description as per MDL ref. MDL-008-11 Rev. 2, dated 21/08/2012; AFMS ref. UAE-AFMS-11-003 Rev. 2, dated 21/08/2012. Note: document reference UAE-CD-E-0004-11 Rev. 1 was modified on 06/09/2012 to record test results; or later revisions of the above listed documents approved by EASA. | None | 07.09.2012 |
| 10014173 | UNIAIR | EASA.A.S.01306 | | | BEECH C90A | Compliance with the B-RNAV requirements for the GPS System Trimble 2000 Approach + | | Master Data List UAE.030.05 Revision 2 dated 21 June 2005. AFMS No. 57/UAE Revision 2 dated 21 June 2005 | 1. This STC is limited to aircraft s/n LJ-1303.2. GPS System Trimble 2000 Approach + installation already approved with IFR privileges. | 20.07.2005 |
| 10014237 | UNIAIR | EASA.A.S.01400 | | | CORVETTE SN 601 | Deviation to SB 34-12 - replacement of ADF in accordance with MDL UAE 010.05 rev 1 dated 11/07/05. Installation of ADF60 according to option 85 aerospace instead of ADF 206. Installation of option 85 on second ADF position instead of first position. (...) | | | 1. This STC is limited to S/N 028 and S/N 0082. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 28.07.2005 |
| 10015036 | UNIAIR | EASA.A.S.03349 | REV. 1 | | MYSTERE FALCON 50 | Installation of 220 VAC Electrical Power Supply and Distribution in Cabin. Note: This STC constitutes the extension of the approval granted under EASA STC ref. EASA.A.S.03349 dated 19 May, 2008, to a second configuration. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Uni Air Enterprise Master Data List MDL-023-07 Revision 5, dated 17th February 2012. Configuration 1: Installation Instructions in accordance with Uni Air Enterprise Service Bulletin SB-UAE-023-07-01 Revision 3, dated 07th March 2008. Uni Air Enterprise (...) | None | 10.04.2012 |
| 10015037 | UNIAIR | EASA.A.S.03350 | REV. 1 | | MYSTERE FALCON 50 | Installation of an In-Flight Passenger Entertainment System in Cabin. Note: This STC constitutes the extension of the approval granted under EASA STC ref. EASA.A.S.03350 dated 19 May, 2008, to a second configuration. | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Uni Air Enterprise Master Data List MDL-022-07 Revision 3, dated 23 January 2012. Configuration 1: Installation Instructions in accordance with Uni Air Enterprise Service Bulletin SB-UAE-022-07-01 Revision 2, dated 18th March 2008. Uni Air Enterprise Airp (...) | STC limited to those Mystere Falcon 50 (MF50) aircraft on which Uni Air STC 10015036 Revision 1 "Installation of 220 VAC Electrical Power Supply and Distribution in Cabin" has been embodied. | 10.04.2012 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------|-------------------|----------------|----------------------|--|---|--|--|---|---------------|
| 10027898 | UNIAIR | P-EASA.A.S.03693 | | | MF900 | Installation of a Flight Data Recorder | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Data List Document MDL-019-07 Revision 5, dated 09th October 2009- Aeroplane Flight Manual Supplement UAE-AFMS-08-006, Revision 2 dated 07th August 2009- Aeroplane Maintenance Manual Supplement UAE-AMMS-08-002, Revision 2 dated 21st September 20 (...) | This design change is applicable on the basic Mystere Falcon 900 aircraft, as per MF900 designation on EASA TCDS A.062 | 13.11.2009 |
| 10031335 | UNIAIR | EASA.A.S.00306 | REV. 1 | | CESSNA CITATION II 550 | Installation of a multi-functions display Honeywell type KMD-850 to display information from the following systems: ACAS II (CAS 67A), EGWPS (MK VIII) and, Weather radar (ART 2000 or RS-181A) | | UAE 015 03 E3 R4, or later approved revisions. | 1. AFM Supplement UAE No 47 issue 1 Revision 1 dated 14 September 2004 or later EASA-approved revision, is a required part of this STC2. Or AFM Supplement UAE No 15 issue 2 revision 2 (for MSN 0609) or revision 5 (for MSN 0190) or later EASA-approved (...) | 02.06.2009 |
| 10042498 | UNIAIR | | | | B300 | Installation of 230 V AC - 50 Hz electrical distribution on aircraft, to allow passengers to plug personal electronic devices. According to SB-UAE-003-12-001. | The Certification Basis of the original product and the following additional airworthiness requirements are applicable to this change. Special Condition "Power supplies for cabin (TGM/25/10)" applied through CRI F-64. The certificated noise and/ or emis (...) | Master data list MDL 003 12 rev.3 dated 28.Nov.2012. Airplane Flight Manual Supplement UAE-AFMS-12-001 rev.1. Airplane Maintenance Manual Supplement UAE-AMMS-12-002 rev.1. | If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. | 05.12.2012 |
| 10015829 | UNIVERSAL AVIONICS SYSTEMS | EASA.IM.A.S.01918 | | FAA STC ST00839LA | BOMBARDIER DHC-8-100 SERIES | Installation of GPS Landing System in de Havilland DHC-8 aircraft (Validation of FAA STC ST00839LA amended 27/02/2006) | | Master Drawing list MDL-2005-1002 Rev CAircraft Flight Manual Supplement AMFS-2005-1002 Rev IR | The approval is limited to Bombardier/de Havilland DHC-8 100 series aircraft | 14.07.2006 |
| 10016206 | UNIVERSAL AVIONICS SYSTEMS | EASA.IM.A.S.02450 | | FAA STC ST01879LA | BOMBARDIER CL-600-2B16 (601-3A) | Universal Avionics Vision-1tm Synthetic System installed in Canadair Challenger CL600 Series. | | Universal Avionics Master Drawing List MDL-2005-1003 Revision C dated 22 May 2006 or later approved revision. Universal Avionics Airplane Flight Manual Supplement AFMS-2005-1003 Revision IR dated 17 May 2006 or later approved revision. | | 23.11.2007 |
| 10016253 | UNIVERSAL AVIONICS SYSTEMS | EASA.IM.A.S.02531 | REV. 2 | FAA STC ST02070 LA | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 | Universal Avionics UNS-1Fw with embedded WAAS. MDL -4 Config (Triple UNS-1Fw FMS with P/N 10709 3 GPS-WAAS/Sirius Antenna) | EASA.IM.A.023 | -EASA STC No. EASA.IM.A.S.02531-FAA STC No. ST02070LA-Universal Avionics MDL No. MDL-2007-1003 Revision 11 dated 11 January 2009 or later approved revision- Universal Avionics AFM Supplement No. AFMS-2007-1005 Revision 00 dated 31 January 2008 or later approved revision | None | 06.07.2009 |
| 10016279 | UNIVERSAL AVIONICS SYSTEMS | EASA.IM.A.S.02568 | | FAA STC ST01763CH-D | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 | Installation of Universal Avionics EF1-890R Flat Panel Displays and MFD-640. | | Universal Avionics Master Drawing List MDL-2007-1001 Revision 00 dated 20 April 2007. Duncan Aviation Airplane Flight Manual Supplement No. 040510022 Revision B dated 20 January 2006? | | 23.11.2007 |
| 10016375 | UNIVERSAL AVIONICS SYSTEMS | EASA.IM.A.S.02698 | | FAA STC ST01640LA | BOMBARDIER CL-600-2B16 (601-3A) | Installation of Universal Avionics Radio Control Units (RCU) | | Universal Avionics Master Document List, Drawing No. MDL-2004-1003, Revision "A", dated May 25, 2004 or later approved revision. Universal Avionics Airplane Flight Manual Supplement Document No. AFMS-2004-1003, Revision IR, dated June 2004, or later approved revision. | Restricted to ASN 5042. | 03.06.2008 |
| 10016614 | UNIVERSAL AVIONICS SYSTEMS | EASA.IM.A.S.03040 | REV. 1 | FAA STC ST02012LA | CL-600-1A11 (CL-600) CL-600-2A12 (CL-601) CL-600-2B16 | Installation of Universal Avionics Application Server Unit (ASU). Minor revision to ASU FAA STC ST02012LA, ESTC EASA.IM.A.S.03040 | EASA.IM.A.023 | -EASA STC No. EASA.IM.A.S.03040-FAA STC ST02012LA-Universal Avionics Systems Master Drawing List, Document No. MDL-2006-1004, Revision E, dated 15 January 2009, or later approved revision. | Restricted to the following MDL Configurations:-3 Application Server Unit (ASU) (ETSO EASA.IM.210.708) with Dual Cursor Control Panels (CCP) (ETSO EASA.IM.210.708) displayed on EF1-890R Navigation Display (ETSO EASA.IM.210.344).-4 Application Server U (...) | 06.07.2009 |
| 10016615 | UNIVERSAL AVIONICS SYSTEMS | EASA.IM.A.S.03041 | | FAA STC ST01303LA | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 | Installation of Universal Avionics UNS-1F/L and optional UniLink communication system | | FAA STC No. ST01303LA dated 25 August 2008. Universal Avionics Systems Master Drawing List, Document No. MDL-8054, Revision G, dated 25 September 2007 or later approved revision. | 1. Restricted to the following configurations:-3 Config: UL-701 with SCN 22-X-4 Config: UNS-1F with SCN 802-X and A702 Maps2. Prior to installation of this modification the installer must determine that the interrelationship between this modification (...) | 07.01.2009 |
| 10016616 | UNIVERSAL AVIONICS SYSTEMS | EASA.IM.A.S.03042 | | FAA STC ST01141LA | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 | Installation of Universal Avionics Terrain Awareness Warning System (TAWS) | | FAA STC No. ST01141LA dated 22 February 2008. Universal Avionics Master Drawing List, MDL-8068, Revision G, dated 20 November 2008, or later approved revision. Airplane Flight Manual Supplement UASC TAWS, Document No. 20346, Revision C or later approved revision. | 1. Restricted to the following configuration:-4 Config: Terrain Awareness and Warning System (TAWS) operating software SCN 11.X (ETSO EASA.IM.210.0170) displayed on MFD-640 (ETSO EASA.IM.210.905)2. Prior to installation of this modification the inst (...) | 07.01.2009 |
| 10045290 | UUDS | | | | A330-200 | Contour A330-200 Business Class seat modification and bumper installation. This modification removes the existing Rockwell Collins IFE equipment, monitors and harnesses from Contour Business class double seats Model S20207-001 and replaces existing Amsafe (...) | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Master Modification Sheet MMS-TMC99-065-25-001-00or later revisions of the above listed documents approved by EASA | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 13.06.2013 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|-------------------|----------------|----------------------|-------------------------------------|--|--|---|---|--------------------------|
| 10014248 | UUDS AERO | EASA.A.S.01470 | REV. 2 | | SEE EASA APPROVED MODEL LIST | Modification of STC EASA.A.S.01470 to allow the baby to sitdown in thebasinet fixed on the partitions or under the hatracks. | CS-25 (JAR 25) | DDP TMC99-025-XX.Static stress analysis report.Flammabilityreport. | None | 03.11.2010 |
| 10034806 | V. KELNER PILATUS CENTRE | | REV. 1 | TCCA STC SA02-80 | PC-12, PC-12/45, PC-12/47 PC-12/47E | Installation Feather Inhibit System | CS 23The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.Do not materially contribute to safety.The certificated noise and/ or emissions I (...) | 1) Aircraft Maintenance Manual Supplement G504071 Rev. B2) Pilot's Operating handbook and approved Airplane Flight ManualSupplement G504091 Rev. Aor later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 20 (...) | None | 18.10.2011 |
| 10039278 | VAN HORN AVIATION, L.L.C. | | | | 206B, L, L1, L3, L4 | Installation of Replacement Tail Rotor Blades | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/ approval.The following paragraphs of FAR 27 amendment 27-44 have been compliedwith:27.51; 27.65 (...) | -206L1-FMS-901 original issue dated 1 December 2011 - Flight ManualSupplement-206L3-FMS-903 revision 1 dated 24 January 2012 - Flight ManualSupplement;-206L4-FMS-904 revision 1 dated 24 January 2012 - Flight ManualSupplement;or later revisions of 1 (...) | When antenna holder PP-10-A601 is installed VNE is limited to 120 kts;When rack PP-10-A421 is installed, carriage of passengers in front seatsis prohibited;When racks PP-10-A401c or PP-A461 are installed, a maximum of 2passengers in rear seats is permitted. | 23.04.2012 |
| 10028237 | VAN RIEMSDIJK ROTTERDAM B.V. | P-EASA.A.S.03898 | | | B747-400F | NXE-0506 Special Cargo Container 125# X 196# with part number 136.020. | The Certification Basis for the original product remains applicable tothis certificate/ approval. | W-<-&->-B Control and Loading Manual supplement RRM510 Issue 2CMM/IPL NXE Cool Container 136.020 doc. Nr. RRM510 issue for later revisions of the above listed documents approved byEASA. | The location of the cargo container is limited to the positionsmentioned in the W-<-&->-B Control and Loading Manual supplementRRM510. | 10.12.2009 |
| 10031360 | VAN RIEMSDIJK ROTTERDAM B.V. | | REV. 3 | | B747-200F/400F/400BCF/8F | Approval of NXE-0506 Special Cargo Container 125" X 196" with PartNumber 136.020 and RZX-0506 Special Cargo Container 125" X 196" withPart Number 136.050 for use in B747 series Freighters | The Certification Basis for the original product remains applicable tothis certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/or emissions levels of the original product areunchanged and remain applicable to this certificate/appoval. | W-<-&->-B Control and Loading Manual Supplement RRM510 Issue 6,B747-400F-W-<-&->-B Control and Loading Manual Supplement RRM520 Issue 2,B747-200F-W-<-&->-B Control and Loading Manual Supplement RRM544 Issue 1, B747-8F-CMM/IPL NXE/RZX CoolContainer (...) | The location of the cargo container is limited to the positionsmentioned in the W-<-&->-B Control and Loading Manual Supplement RRM510(B747-400F/400BCF)-<-&->- RRM520 (B747-200F) and RRM544 (B747-8F) | 08.08.2012 |
| 10032842 | VAN RIEMSDIJK ROTTERDAM B.V. | | REV. 1 | | B747-200F, 400F | RGX-0517 Special Cargo Container 238,5" X 96" with Part number136.080-000 | The Certification Basis for the original product remains applicable tothis certificate/ approval. | Boeing 747 Freighter W-<-&->-B Manual Pallet size code GCMM/IPL RGX Cool Container Ph. 136.080-xxx doc. No. 25-55-17 / RRM517Issue 3or later revisions of the above listed documents approved by EASA | The loading limitations of the RGX container are according to thelimitations in the applicable W-<-&->-B manuals for the NAS3610 SizeCode G ULD's | 24.02.2011 |
| 10015733 | VENTURA AEROSPACE, Inc. | EASA.IM.A.S.01755 | REV 1 | FAA STC ST01806LA | BOEING 737-400 | Installation of a 9g Cargo Barrier according to FAA STC ST01806LARevision to existing EASA STC EASA.IM.A.S.01755 (737-400 added). | | - Master Data List #Boeing 737-200 / 300 9g Cargo Barrier Installation# Report No. 3168 Rev.E dated 21 June 2005 or later approvedrevision.- Instructions for Continued Airworthiness #Boeing 737 9g Cargo Barrier Installation# Report No. 2210, Rev. B (...) | 1) Prior to installation of this modification the installer must determine that the interrelationship between this modification andany other previously installed modification willintroduce no adverse effect upon the airworthiness of the product. The (...) | 28.05.2008 |
| 10038866 | VENTURA AEROSPACE, Inc. | | | FAA STC ST02115LA | B757-200 | Installation of 9g Forward Cargo Barrier on the Main Deck | The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate.For areas for areas of change the Certification Basis is CS25 Amendment2.Requirements at amendme (...) | Master Data List 2686, revision F, 08/01/2008or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions ofthis decision) | To be installed on a/c converted to a Combination Passenger/ Freightby ST Aerospace Inc, EASA STC 10038822. | 20.03.2012 |
| 10016125 | VIDEO TECHNOLOGY SERVICES, Inc. | EASA.IM.A.S.02318 | | FAA STC ST01998NY | BOEING 767-300 | 1. Configuration 1: Installation of LCD Monitor System2. Configuration 2: Installation of LCD Monitor System | | 1. Configuration 1: in accordance with Video Technology Services, Inc. Technical Drawing List Document No. 205-D-334, Revision B, dated August 9, 2004.2. Configuration 2: in accordance with Video Technology Services, Inc. Technical Drawing List Doc (...) | 1. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall gvethe other person written evidence of that permission.2. The aircraft must be maintained in accordance with Video Technology Ser (...) | 05.06.2007 |
| 10027220 | VIDEO TECHNOLOGY SERVICES, Inc. | | | FAA STC ST02720NY | B767-200, 300 | Installation of Digital Video Player (DVP) on a B767 a/c | Certification Basis in accordance with EASA TCDS IM.A.035 plus-CRI H-01/Special Condition H-01 "Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EIWIS"The certificated noise and/ or emissions levels of the original productare unchanged (...) | - Video Technology Services, Inc. Master Technical Drawing List Document# 205-D-400, Revision C dated July 9, 2009- Video Technology Services, Inc. Instructions for ContinuedAirworthiness Document # 205-D-417, dated March 15, 2009- VideoTechnology (...) | None | 15.09.2009 |
| 10036188 | VIDEO TECHNOLOGY SERVICES, Inc. | | | FAA STC ST03006NY | A340-200/300/600 | Title: LCD Monitor SystemsDescription: Installation of LCD Monitors and Digital Player replacingCRT Monitors and Cassette Recorders in the same locations using the samemounting provisions, cables and connectors.Validation of FAA ST03006NY dated August 4, 2011. | The Certification Basis for the original product remains applicable tothis certificate/ approval. | Associated documents of Video Technology Services, Inc.- Cert plan 205-D460 Rev B. Dated 20-5-2011- Master Technical List document 205-D-463 Rev B, dated 25-5-2011- Instructions for Continued Airworthiness 205-D-467 Rev. -dated2-3-2011 | None. | 18.08.2011 |
| 10042226 | VIDEO TECHNOLOGY SERVICES, Inc. | EASA.IM.A.S.01071 | | FAA STC ST01089NY | 767-200, 767-300 | | | Technical Drawing List 205-D-480, Revision -30/03/2012;ICA Document 205-D-481, Revision B, 15/07/2012;or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions ofthis d (...) | Prior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce roadverse effect upon the airworthiness ofthe produc (...) | 23.03.2005 31.01.2013 |
| 10043481 | VIDEO TECHNOLOGY SERVICES, Inc. | | | FAA ST03223NY | B767-200/- 300 | Installation of LCD Monitors. | | | | |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|-------------------|----------------|--------------------------|--|---|---|--|---|---------------|
| 10030509 | VIH AEROSPACE Inc. | | | TCCA STC SH09-54 | S92A | Triple Stacking Litter System Installation/Validation of TCCA STC SH09-54 Issue 1 | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -MS08036-01 revision 1 - DECA Aviation Modification Summary-FMS08036 revision 2 dated 19 November 2009 - Flight Manual Supplement-; CH192-0308-01-2520-ICA revision 2 dated 24 December 2009 - Cougar Helicopters Instructions for Continued Airworthiness (...) | None | 21.06.2010 |
| 10040358 | VIH AEROSPACE Inc. | | REV. 1 | TCCA STC SH08-29 | S-92A | LHS Crew Bubble Window Installation | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -VS92-0803-FMS revision 1 dated 18 April 2012 - Flight Manual Supplement or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision)-VS92-0803-MDL revision 6 (...) | None | 08.08.2012 |
| 10043045 | VIH AEROSPACE Inc. | EASA.IM.R.S.01476 | | TCCA STC SH08-29 | S92A | | HS Crew Bubble Window Installation | - VS92-0803-MDL revision 0 dated 16 May 2008 or later EASA approved revision - Master Drawing List- VS92-0803-FMS revision 0 dated 16 May 2008 or later EASA approved revision - Flight manual Supplement- VS92-0803-ICA revision 0 dated 16 May 2008 or (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. This Cer (...) | 25.02.2009 |
| 10043046 | VIH AEROSPACE Inc. | EASA.IM.R.S.01455 | | TCCA STC SH08-10 ISSUE 1 | S-92A | Passenger Seating Configuration. | | 1. VS92-371-PLN revision 1 or later approved revision - Drawing. 2. VS92-371-PLN-FMS revision 2 dated 28 January 2008 or later approved revision - Flight manual Supplement. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The inst (...) | 01.08.2008 |
| 10043047 | VIH AEROSPACE Inc. | EASA.IM.R.S.01288 | | TCCA STC NO. SH06-41 | S-92A | Sea Tray Installation. | | Installed in accordance with VIH Aerospace Master Drawing List VS92-0608-MDL, Revision 0, or later approved revisions. Maintenance in accordance with VIH Aerospace Instructions for Continued Airworthiness VS92-0608-ICA, Revision 0, or later approved revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined (...) | 10.05.2007 |
| 10043048 | VIH AEROSPACE Inc. | EASA.IM.R.S.01282 | | TCCA STC SH06-36 | S-92A | Bubble Window Installation. | | Installed in accordance with VIH Aerospace Master Drawing List VS92-0611-MDL, Revision 0, or later approved revisions. Maintenance in accordance with VIH Aerospace Instructions for Continued Airworthiness VS92-0611-ICA, Revision 0, or later approved revisions. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined (...) | 10.05.2007 |
| 10032527 | VIKING AIR LIMITED | | | TCCA STC SA10-69 | DHC-6 SERIES 400 | Installation of RW Martin Air Conditioning System in accordance with TCCA STC SA10-69. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Title 14 CFR Part 23 Amendment 23-0 23.303, 23.307, 23.601, 23.609, 23.1367 Amendment (...) | Flight Manual Supplement AC-64-1A, Initial issue, TCCA approved September 9, 2010 Instruction of Continued Airworthiness AC-64-11, Initial issue, dated August 4, 2010 R.W. Martin Inc. Drawing List, Report ER218-2502V, Revision 3, dated September 7, 2010 | None | 12.11.2010 |
| 10015333 | VISIONSAFE CORPORATION | EASA.IM.A.S.00354 | REV. 7 | FAA STC ST00892LA | SEE EASA APPROVED MODEL LIST | Installation of Emergency Vision Assurance System (EVAS TM). Validation of FAA STC ST00892LA as amended on May 29, 2012. | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. FAR requirements FAR 25.853(a) and FAR 25.1713(c) at amendment level 25-123. The requirem (...) | Master Drawing List 404, Revision "I", April 24, 2012 Drawing List 110, Revision "T#", May 31, 2012 Airplane Flight Manual Supplement 8999-404, Revision "Original", May 22, 2012. ICA document 8017, Revision "10" or later revisions of the above listed docum (...) | None | 15.01.2013 |
| 10015744 | VISIONSAFE CORPORATION | EASA.IM.A.S.01774 | | FAA STC ST01012W1 | BOMBARDIER CL-600-1A11 (CL-600) BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-2B16 BOMBARDIER CL-600-2B19 | Installation of Emergency Vision Assurance System (EVAS TM) | | 1) Midcoast MDL 34-416G001, Rev. B, or Vision Safe Corporation MDL 176, Rev. C or later FAA approved revisions 2) AFMS to Canadair CL-600 AFM STC No ST01012W1(3) FAA STC ST01012W1 | 1) Certification Basis: applicable requirements of a. r. TCDSs2) This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 22.12.2005 |
| 10016007 | VISIONSAFE CORPORATION | EASA.IM.A.S.02172 | REV. 1 | FAA STC ST01057W1 | BD-700-1A10, BD-700-1A11 | Installation of Emergency Vision Assurance System (EVAS) Model: 107STC-191 and 107STC-192 | | -MDL: Midcoast Aviation 35-416G001 Rev A or Vision Safe No.206 Rev H-AFM-Supplement: Midcoast Aviation 35-416M001 Rev Original or Vision Safe No. 1, Rev 1 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decis (...) | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product (...) | 10.02.2009 |
| 10016045 | VISIONSAFE CORPORATION | EASA.IM.A.S.02216 | | FAA STC ST00892LA | AIRBUS A330-201 AIRBUS A330-202 AIRBUS A330-203 AIRBUS A330-223 AIRBUS A330-243 AIRBUS A330-301 AIRBUS A330-321 AIRBUS A330-322 AIRBUS A330-323 AIRBUS A330-341 AIRBUS A330-342 AIRBUS A330-343 | Installation of Emergency Vision Assurance System (EVAS TM) / EASA Validation of FAA approved Supplemental Type Certificate ST00892LA. 1. Boeing 737-600, -700, -700C, -800 and -900; Airbus 330-201, -202, -203, -223, 243; Airbus 330-301, -321, 322, -323, -341, -342, -343 models | | 2. FAA STC ST00892LA, EVAS Compliance Summary for FAA Part 25 and EASA CS 25 dated March 2006, Master Drawing List Number 222 for Vision Safe Corporation - EVAS Model 107STC-215 and 107STC-216 - Airbus A330 Series and A340 Series, Revision G (...) | 2. Vision Safe Corporation Instructions for Continued Airworthiness, Rev. 04, dated March 2006. | 23.07.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|------------------------|---------------------------|----------------|----------------------|---|---|--|--|---|---------------|
| 10016155 | VISIONSAFE CORPORATION | EASA.M.A.S.02357 | | FAA STC ST00731LA | BOEING 767-200 BOEING 767-300 SERIES | Installation of Emergency Vision Assurance System (EVAS) in accordance with FAA Approved Vision Safe Corporation Master Drawing List Number 144, Revision #C dated June 9, 1999, or later FAA Approved Revisions, Vision Safe Corporation Airplane Flight (...) | | see above | This installation should not be incorporated in any aircraft unless it is determined that the interrelationship between this installation and any previously approved configuration will not introduce any adverse effect upon the airworthiness of the air (...) | 30.03.2007 |
| 10030483 | VISIONSAFE CORPORATION | | | FAA STC ST00892LA | BD100-1A10 | Installation of Emergency Vision Assurance System (EVAS) Model 107STC-235 and 107STC-236 on Bombardier BD-100-1A10 EASA Validation of FAA approved Supplemental Type Certificate ST00892LA | The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | FAA STC ST00892LA with Approved Model List. See AML page 11 of 13 for BD-100-1A10. Master Drawing List Number 226 for Vision Safe Corporation EVAS Model 107STC-235 and 107STC-236 on Bombardier Aerospace BD-100-1A10, Revision C dated 14th March 2007-BD-10 (...) | None | 17.06.2010 |
| 10043091 | VITROCISET S.p.A. | EASA.A.S.00411 | REV. 2 | | 500, 550 SERIES | Installation of Honeywell EGPWS. Revision of EASA.A.S.00411 due to extension of applicability. Note: This STC supersedes EASA.A.S.00411 Revision 1. | | Master Drawing List DRW No 700-1004-000 Rev. 1, dated 23 December 2004. | none | 07.01.2009 |
| 10043092 | VITROCISET S.p.A. | EASA.A.S.00412 | REV. 2 | | 500, 550 SERIES | Honeywell ART 2000/2100 Radar System Installation. Revision of EASA.A.S.00412 due to extension of applicability. Note: This STC supersedes EASA.A.S.00412 Revision 1. | | Master Drawing List DRW No 400-0804-000 Rev. 1 dated 23/12/2004 and subsequent approved revision. | none | 07.01.2009 |
| 10043097 | VITROCISET S.p.A. | EASA.A.S.0061 | REV. 1 | | F27 | STC No. 02/04; Electronic flight Instrument System (EFI 550) and Vertical Gyros Installation. Note: This STC supersedes EASA.A.S.0061. | | none | 1. „Instruction for Continued Airworthiness 400-0204-600-000 Rev. IR dated 30.04.2004 Aircraft Flight Manual Supplement No. 400-0204-800-000 Rev. IR dated 30.04.2004. This STC is approved only for the product configuration as defined in the approved des (...) | 07.01.2009 |
| 10043098 | VITROCISET S.p.A. | EASA.A.S.00760 | | | F27 MK 500 | Traffic Collision Avoidance Systems (TCAS II) installation. Note: This STC supersedes EASA.A.S.00761. | | Master Drawing List No 600-2004-000-000 Rev. IR dated 30 March 2005. | 1. Airplane Flight Manual Supplement No 600-2004-800-000 Rev. IR dated 25 January 2005. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other (...) | 07.01.2009 |
| 10043099 | VITROCISET S.p.A. | EASA.A.S.00761 | REV. 1 | | F27 MK 500 | Traffic Collision Avoidance Systems (TCAS II) installation. Note: This STC supersedes EASA.A.S.00761. | | Master Drawing List No 600-2004-000-000 Rev. IR dated 30 March 2005. | 1. Airplane Flight Manual Supplement No 600-2004-800-000 Rev. IR dated 25 January 2005. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other (...) | 07.01.2009 |
| 10043100 | VITROCISET S.p.A. | EASA.A.S.00762 | | | F27 MK 500 | Radar System Replacement with Bendix King ART 2000 System. Note: This STC supersedes EASA.A.S.00762. | | Master Drawing List No 400-0303-000-000 Rev. IR dated 29 July 2003 or later approved revisions. | 1. Airplane Flight Manual Supplement No 400-0303-800-000 Rev. IR dated 4 June 2003. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other ai (...) | 07.01.2009 |
| 10043122 | VITROCISET S.p.A. | EASA.A.S.02993 REVISION 1 | REV. 1 | EASA.A.S.02993 | 500, 550 SERIES | Installation of TAWS/RMI Sandel ST3440 Class A. Note: This STC supersedes EASA.A.S.02993. | | AFM Supplement DRW No. 34-97-0105-0800-000 Rev. IR dated 20.12.2006 and subsequent approved revisions. Master Drawing List DRW No. 34-97-0105-0000-000 Rev. IR dated 26.02.2007 and subsequent approved revisions. | This STC is approved only for the product configuration as defined in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined (...) | 07.01.2009 |
| 10043124 | VITROCISET S.p.A. | EASA.A.S.03065 REVISION 1 | REV. 1 | EASA.A.S.03065 | SA 227 | GPWS MKVI/VIII replacement with EGPWS MKVI/MKVIII Honeywell. Note: This STC supersedes EASA.A.S.03065. | | AFM Supplement DRW No. 34-40-0706-0800-000 Rev. IR dated 09.02.2007 and subsequent approved revisions. Master Drawing List DRW No. 34-40-0706-0000-000 Rev. IR dated 16.02.2007 and subsequent approved revisions. | This STC is approved only for the product configuration as defined in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with other aircraft/engine configurations shall be determined (...) | 07.01.2009 |
| 10043127 | VITROCISET S.p.A. | EASA.A.S.03106 REVISION 2 | REV. 2 | EASA.A.S.03106 | MU-300 | Installation of Honeywell MKVI EGPWS. Note: This STC supersedes EASA.A.S.03106 Revision 1. | | AFM Supplement DRW No. 34-97-0407-0800-000 Rev. IR and subsequent approved revisions. Master Drawing List DRW No. 34-97-0407-0000-000 Rev. IR dated 2 July 2007 and subsequent approved revisions. | 1. Least B.M. No. 15/07 must be applied before the installation. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will not introduce (...) | |
| 10043130 | VITROCISET S.p.A. | EASA.A.S.03340 REVISION 1 | REV. 1 | EASA.A.S.03340 | 550 | Installation of Rockwell Collins Mode S Transponder System. Note: This STC supersedes EASA.A.S.03340. | | AFM Supplement DRW No. 34-54-1707-800-000 Rev. IR dated 30-07-2007 and subsequent approved revisions. MDL DRW No. 34-54-1707-000-000 Rev. IR dated 25-10-2007 and subsequent approved revisions. | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will not introduce any adverse effect upon the airworthiness of the product. The inst (...) | 07.01.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|--------------------------|-------------------|----------------|---------------------------|---|---|---|--|--|---------------|
| 10043141 | VITROCISSET S.p.A. | EASA.A.S.03780 | | ENAC STC CS-45 | 500,550 | Installation of Honeywell GPWS Mark VI in accordance with MDL600-1700-000-000 dated Jan. 8th 2001 or later EASA approved revisions. Note 1. This STC reflects the transfer of ENAC STC CS-45 from LEAT S.r.l. to VITROCISSET S.p.A.2. This STC supersedes ENAC STC CS-45. | | None | AFM Supplement Doc. N. 600-1700-800-000 dated Jan. 8th 2001 or later EASA approved revisions. This Certificate shall remain valid unless otherwise surrendered or revoked. | 07.01.2009 |
| 10043142 | VITROCISSET S.p.A. | EASA.A.S.03781 | | ENAC STC CS-060 | 402B | Ammodernamento avionica ed installazione GPS KLN 94 per operazioni BRNAV. Note 1. This STC reflects the transfer of ENAC STC CS-060 from LEAT S.r.l. to VITROCISSET S.p.A.2. This STC supersedes ENAC STC CS-060. | | Doc. DRW 400-0901-000 e successive revisioni approvate. | Aircraft Flight Manual Supplement (Allegato N) 1st Edition 16 July 2001 e successive revisioni approvate. This Certificate shall remain valid unless otherwise surrendered or revoked. | 07.01.2009 |
| 10043143 | VITROCISSET S.p.A. | EASA.A.S.03782 | | ENAC STC CS-071 | ENAC STC CS-071 | Installation of EGPWS Universal Avionics Systems with GPS 1000 and MFD640 (> <) - Note 1. This STC reflects the transfer of ENAC STC CS-071 from LEAT S.r.l. to VITROCISSET S.p.A.2. This STC supersedes ENAC STC CS-071. | | Master Drawing List No. 700-0202-000-000 IR dated 17.06.2002 and later approved revisions. | Included in the Supplemental Flight Manual No. 700-0202-800-000 Rev. IR dated 17.06.2002 and later approved revisions. This Certificate shall remain valid unless otherwise surrendered or revoked. | 07.01.2009 |
| 10043144 | VITROCISSET S.p.A. | EASA.A.S.03783 | | ENAC STC CS-070 | 550 CITATION II | Upgrade FDR installation with installation of Digital SSFDR type F1000. Note 1. This STC reflects the transfer of ENAC STC CS-070 from LEAT S.r.l. to VITROCISSET S.p.A.2. This STC supersedes ENAC STC CS-070. | | Master Drawing List No. 900-0402-000-000 IR dated 17.06.2002 and later approved revisions. | Included in the Supplemental Flight Manual No. 900-0402-800-000 Rev. IR dated 17.06.2002 and later approved revisions. This Certificate shall remain valid unless otherwise surrendered or revoked. | 07.01.2009 |
| 10043145 | VITROCISSET S.p.A. | EASA.A.S.03784 | | ENAC STC CS-062 | 550 | Coupling of GPS Trimble 2101 I/O Approach Plus with Autopilot SPZ 500. Note 1. This STC reflects the transfer of ENAC STC CS-062 from LEAT S.r.l. to VITROCISSET S.p.A.2. This STC supersedes ENAC STC CS-062. | | MDL DWG No. 100-0502-000-000 dated 24.04.2002 and later approved revisions. | Included in the Supplemental Flight Manual No. 100-0502-000-000 dated 25.03.2002 and later approved revisions. This Certificate shall remain valid unless otherwise surrendered or revoked. | 07.01.2009 |
| 10043146 | VITROCISSET S.p.A. | EASA.A.S.03785 | | ENAC STC CS-091 | SA 227 | Coupling of GPS GARMIN 155XL with Autopilot. Note 1. This STC reflects the transfer of ENAC STC CS-091 from LEAT S.r.l. to VITROCISSET S.p.A.2. This STC supersedes ENAC STC CS-091. | | Doc. No. 100-0203-000-000 Master Drawing List dated 07.07.2003 and later approved revisions. | Included in the Supplemental Flight Manual No. 100-0203-800-000 dated 13.03.2003 and later approved revisions. This Certificate shall remain valid unless otherwise surrendered or revoked. | 07.01.2009 |
| 10027778 | V-PLANE ENGINEERING GmbH | P-EASA.A.S.03977 | | | PC-12, PC-12/45, PC-12/47, PC-12/47E | Heavy Duty Floor | The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | -Engineering Order BUPC12-EO-0001-01-02 Issue 02-Aircraft Flight Manual Supplement BUPC12-AM-5300-01-02 Issued later revisions of the above listed documents approved by EASA | Heavy Duty Floor P/N 0127258 is limited to from S/N 101 to S/N 999 except S/N 545 with configuration in accordance with Bucher CMM E014-16-01 Heavy Duty Floor P/N 0127470 and P/N 0127470-401 is limited to from S/N 1001 on which configuration in accordance with Bucher CMM E016-19-08 | 03.11.2009 |
| 10028296 | V-PLANE ENGINEERING GmbH | P-EASA.A.S.03978 | | | ERJ170-100STD, LR ERJ170-200STD, LR | Installation of a removable stretcher | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: EASA CS-25 Amendment 6 The certificated noise and/or emissions levels of the original product (...) | Engineering Order: BUE170-EO-0001-01-01, Issue 01 Master Drawing List: BUE170-CS-2520-05-01, Issue 01 Installation Instructions: BUE170-TO-0001-01-01, Issue 01 Continued Airworthiness Document: BUE170-CS-2520-02-01, Issue 01 User Manual: BUE170-CS-2520- (...) | None | 16.12.2009 |
| 10043290 | V-PLANE ENGINEERING GmbH | EASA.R.S.01538 | | | BELL 206B | Avionics Systems Exchange/Installation - CNI Equipment# „Installation of Dual Garmin GNS430AW Com. NAV, GPS with MidContinent MD41-1488A annunciator/switching units and Garmin GAD42 instrument interfaces, using the existing HSI (relocated to L/R sid (...) | | „AA1327-02-01R3 dated 06 March 2009 # Technical order (contains installation instructions); „HP2061-AM-2350-01-01 issue 2 dated 16 March 2009 # Flight Manual Supplement GMA340H; „HP2061-AM-3010-01-01 issue 2 dated 16 March 2009 # Flight Manual Supp (...) | none | 16.06.2009 |
| 10030059 | West Pacific Air, LLC | EASA.IM.A.S.03276 | | FAA STC SA01156SE | B36TC | Installation of a Pratt and Whitney PT6A-21 engine flat rated to 500HP at 2,200 RPM, and Hartzell model HC-E4N-3N/D8292 and HC-E4N-3N/D8292(B,K)-2 propellers. This certificate has been first issued on 17 May 2010 but has been re-issued for administrative reasons due to the loss of the original. | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certificated noise and/or emissions levels. | Installation in accordance with Rocket Engineering Master Drawing List 450.00.000, Revision S, dated June 30, 2007, or later approved revisions. Operation in accordance with Turbine Air/West Pacific Air LLC Flight Manual Supplement ref 450.1001, Change 9 (...) | Model B36TC Only The MT propeller option on the original FAA STC is not EASA approved at this time. In order to install the MT propeller compliance with EU noise and environmental standards must be shown. | 24.04.2012 |
| 10015916 | WESTERN AVIONICS | EASA.IM.A.S.02048 | | TCCA STC SA04-26 ISSUE 3 | FAIRCHILD SWEARINGEN SA 228 SE FAIRCHILD SWEARINGEN SA 227 SE | Installation of Honeywell EGPWS Mark VI/VIII computer, air data computer, circuit breakers, annunciators, flap position switch, OAT probe, GPS antenna, cockpit speaker and associated wiring according Transport Canada Supplemental Type Certification SA04-26, Issue 3, March 09, 2005 | | 1. Installation is to be done in accordance with Transport Canada approved Western Avionics Engineering Order 92350-401, Revision 4, dated 05 January 2005, or later Transport Canada approved revision. 2. Transport Canada approved Western Avionics FI (...) | | 23.10.2006 |
| 10016041 | WESTERN AVIONICS | EASA.IM.A.S.02210 | | TCCA STC SA01-155 ISSUE 5 | DE HAVILLAND CANADA DHC6-300 | Installation or Upgrade of a Honeywell MK VI Enhanced Ground Proximity Warning System (EGPWS) | | For installation: Transport Canada approved Western Avionics Engineering Order No. 91429-401, Revision 2, dated 16-Mar-2006, or later approved issue For Upgrade: Transport Canada approved Western Avionics Engineering Order No. 91429- (...) | Approval is limited to the CIC Air Data Module. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs "Description" and "Associated Technical Documentation". Compatibility with (...) | 12.01.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|---------------------|----------------|--------------------------------|---|---|---|---|---|---------------|
| 10016683 | WESTERN AVIONICS | EASA.IM.A.S.C.01002 | | TCCA STC SA03-13; EASA.IM.A.S. | BOMBARDIER DHC-8-100 SERIES BOMBARDIER DHC-8-102 BOMBARDIER DHC-8-103 BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-300 SERIES BOMBARDIER DHC-8-311 BOMBARDIER DHC-8-315 | Installation (Upgrade) of a Single or Dual Universal Avionics UNS-1C+ or UNS-1E Flight Management System (FMS) | | -Top Document List 91846-400, Revision 5 dated 29 September 2004 Western Avionics AFM Supplement 71967-803, Revision 1 dated 05 August 2003 (EMS) or 71967-804, Initial Issue dated 05 August 2003 (EPIS), (AFM Supplement 71967, Revision 1 dated 05 Augu (...) | 1. Installation (Upgrade) of the Single or Dual Universal Avionics UNS-1+ or UNS-1E Flight Management System (FMS) is to be done in accordance with Transport Canada approved Western Avionics Engineering Orders 71967-401 Revision 5 dated 17 September (...) | 20.05.2005 |
| 10028593 | WESTERN AVIONICS | | | TCCA STC SA09-65 | DHC-8-102, -103, -106 DHC-8-201, -202 DHC-8-301, -311, -314, -315 | Standby Altimeter Installation TCCA STC SA09-65, dated 07 August 2009 | FAR-25 as per EASA.IM.A.191 The Certification Basis for the original product remains applicable to this certificate/ approval. The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Installation: Must be in accordance with Western Avionics Engineering Order 119975-401, Initial Issue, TCCA approved 07 August 2009. Operation: Must be in accordance with Western Avionics Flight Manual Supplement 119975-801, Initial Issue, TCCA approved 0 (...) | None | 27.01.2010 |
| 10029332 | WESTERN AVIONICS | | REV. 1 | TCCA STC SA03-53 ISSUE 1 | DHC-8-102, -103, -106, DHC-8-201, -202, -301, DHC-8-311, -314, -315 | Primus 800 Weather Radar Installation TCCA STC SA03-53, Issue 1, dated 28.05.2003. Removal of the limitation for instructions for Continued Airworthiness Electrical Wiring Interconnection System - EWIS ICA - in accordance with Special Condition H-01 (...) | FAR-25 as per EASA.IM.A.191 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Special Condition H-01 "Enhanced Airworthiness Programme for (...) | Installation must be in accordance with Western Avionics Engineering Order 90404-401, Rev. 2. Maintenance must be in accordance with Western Avionics Maintenance Manual Supplement 90404-701, Rev. 2 or later revisions of the above listed documents approved (...) | A previously approved Weather Radar System installation is a prerequisite for the installation. | 30.11.2010 |
| 10030455 | WESTERN AVIONICS | | | TCCA STC SA03-42 | DHC-8-102, DHC-8-103, DHC-8-106, DHC-8-201, DHC-8-202, DHC-8-301, DHC-8-302 | Collins Proline II Installation TCCA STC SA03-42, Is. 1, dated 8 May 2003 | FAR-25 as per EASA.IM.A.191 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Special Condition H-01 "Enhanced Airworthiness Programme for (...) | Installation must be in accordance with Western Avionics Engineering Order 89596-401, Rev. 1, dated April 24, 2003, or later Transport Canada approved revision. Maintenance must be in accordance with Western Avionics Maintenance Manual Supplement, 89596 (...) | None | 16.06.2010 |
| 10030955 | WESTERN AVIONICS | | | TCCA STC SA04-96 | DHC-8-102/-103/-106 DHC-8-201/-202 DHC-8-301/-302 | Second ADF Installation TCCA STC SA04-96 Issue 1 dated 23 December 2004 | FAR 25 as per EASA.IM.A.191 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product remain applicable to this certificate/ approval. | Installation of Second Collins ADF-60A or Honeywell (Bendix/King)/KDF-806 ADF Systems must be in accordance with TCCA approved Western Avionics Engineering Order 99459-401, original issue dated 31 August 2004. Maintenance must be in accordance with Bomber (...) | None | 16.07.2010 |
| 10037518 | WESTERN AVIONICS | | | | DHC-6-1/-100/-200/-300 | Installation of Garmin GPS 155XL for supplemental IFR Navigation in acc. with TCCA STC SA01-93, Issue 1 | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. FAR 23, Amdt. 23-0-23.1, 23.301, 23.303, 23.305, 23.307, 23.321, 23.561, 23.601, 23.60 (...) | Western Avionics Modification data Summary 71961-900, Revision 2, dated May 31, 2001. Western Avionics Flight Manual Supplement 71961-801, Initial Issue, TCCA approved August 1, 2001 or later revisions of the above listed documents approved by EASA in acc (...) | None | 02.12.2011 |
| 10040641 | WESTERN AVIONICS | | | TCCA STC SA10-104 | DHC-8-102, DHC-8-103, DHC-8-106 DHC-8-201, DHC-8-202, DHC-8-301 DHC-8-311, DHC-8-314, DHC-8-315 | Installation of a Western Avionics PicoCube Communications System TCCA STC SA10-104, Issue 2 | FAR 25 as per EASA.IM.A.191, Issue 5 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions level (...) | Installation: Must be in accordance with Western Avionics Engineering Order 120760-401, Revision 1, Transport Canada approved 2010 December 15. Operation: Must be in accordance with approved Western Avionics Flight Manual Supplement 120760-802, IR, Transp (...) | Structural and antenna provision in accordance with the latest applicable issue of EASA STC MOT.21NE2.139 are required. | 16.07.2012 |
| 10043076 | WESTERN AVIONICS | EASA.IM.A.S.03116 | | FAA STC SA08-82 | DHC-8-102,-103,-106 | Replacement of the existing Honeywell AH-600 Attitude and Heading Reference System (AHRS) with the installation of a SFIM industries F227 Aircraft Piloting Interior Reference Sensor (APIRS), ref. TCCA SA08-82, Issue 1, dated 24 November 2008 | | - Installation in accordance with TCCA approved WA Engineering Order 117334-401, dated 17 September 2008. Operation in accordance with TCCA approved WA Flight Manual Supplement 117334-801, dated 24 November 2008. Maintenance in accordance with TCCA ac (...) | none | 18.05.2009 |
| 10043355 | WESTERN AVIONICS | | | TCCA STC SH11-20 ISSUE 1 | R44 & R44 II | Installation of Aspen Avionics EFD 1000H/500H Electronic Flight Display Systems in accordance with Western Avionics Engineering Order CET3-401. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: FAR 27.610 at Amdt. 37. The requirements for environmental protection and the associated certified noise and/ or emissions level (...) | Flight Manual Supplements: Configuration, ..., ..., Supplement, ..., Approval Date EFD1000H PFD, ..., CE13-801, ..., 2011 June 23 EFD1000H PFD & EFD500H MFD, CE13-802, ..., 2011 June 23 EFD1000H PFD & EFD500H MFD, CE13-803, ..., 2011 June 23, or (...) | none | 23.01.2013 |
| 10016705 | WESTERN PROPELLER (PACIFIC) Ltd. | EASA.IM.P.S.01003 | | TCCA STC SP96-1 | | Installation of a Zero Stop Latch Assembly To hold the propeller in zero pitch during engine start to prevent float and ski equipped aircraft from surging forward during transition from feather to beta range. In accordance with the data contained in W (...) | | - Transport Canada STC No. SP96-1, Issue 3 dated 14.4.08. FAA STC No. SP00519NY dated 9.10.1996. FAA TCDS No. P60GL, Rev. 3 dated 30.5.06. WPL-C703-6 Zero Latch Assembly, Rev. B dated 7 May 2001 incl. - WPL-C703-1 Zero Stop Latch Assembly, Rev. (...) | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the aircraft (...) | 23.01.2009 |
| 10029049 | WESTERN PROPELLER (PACIFIC) Ltd. | EASA.IM.A.S.02907 | | FAA TC A37CE | CESSNA 208A, 208B | Installation of a modified McCauley 3GFR34C703 Propeller incorporating a zero Latch System. | CS-23.B NSC | Installation in accordance with Cessna Manual P/N D2077-13. Operation must be in accordance with Western Propeller Company Ltd. Aircraft Flight Manual Supplement WPL-C703-5 Re. A, dated 7th May 2001 or later approved revision. | none | 23.07.2009 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|---------------------------------|---------------------|----------------|----------------------|--|---|--|--|---|---------------|
| 10015762 | WIGGINS AIRWAYS, INC. | EASA.IM.A.S.01798 | | FAA STC ST00052BO | RAYTHEON 1900D | Validation of FAA STC ST00052BO Incorporation of a RockwellCollins TCAS 94 System (TCAS II) with TTR-921 Change 7.0 R/T, dual TUI-920 TARA/VS1 Indicators dual TDR-94D Mode S Transponders, CTL-92T TCAS Control and CTL-92 Transponder Control | | Wiggins Airways, Inc. Engineering report No. 7065, Revision02 dated November 5, 2000, or later approved revisions.AFM:Wiggins Airways, Inc. Airplane Flight Manual Supplement ref. No 227 Revision 02, dated March 30, 2006 or later approved revision.Maintenance Manual: Rockwell Collins Manual for tra | This STC does not comply with requirements for Mode S elementary surveillance or for Mode S enhanced surveillance.Compatibility of this design change with other previously approved modifications must be determined by the installer | 05.04.2006 |
| 10015352 | WILCO INC | EASA.IM.A.S.00598 | | FAA STC SA00741W1 | BEECH 100 BEECH 1900 BEECH 200 BEECH 200C BEECH 200T BEECH 300 BEECH 300LW BEECH 65-90 BEECH 65-A90 BEECH 99 BEECH 99A BEECH A100 BEECH B100 BEECH B200 BEECH B200C BEECH B200T BEECH B300 BEECH B90 BEECH B99 BEECH C90 BEECH C90A BEECH C99 BEECH E90 BEECH F90 RAYTHEON 1900C RAYTHEON 1900D | Installation of the Concorde RG-380 E/K Series VRSLA Batteries, as per FAA STC SA00741W1. | | - Installation Instructions W3556005. | | 02.03.2005 |
| 10015378 | WILKINSON AIRCRAFT SERVICES INC | EASA.IM.A.S.01055 | | FAA STC SA10107SC | PA-46-310P | Installation of a 2 direct drive vacuum pump using the installation instruction document number 0001, revision Original dated 2/2/02 or later FAA approved revision | | -FAA STC SA10107SC issued November 12, 2002-Installation Instruction Document number 0001, revision Original dated 2/2/02 or later FAA approved revision-Airplane Flight Manual Supplement document number 0001, revision Original dated 2/2/02 or later FAA approved revision | 1. Airplane Flight Manual Supplement document number 0001,revision Original dated 2/2/02 or later FAA approved revision required2. This STC is approved only for the product configuration as defined in the approved design data referred to in the para (...) | 10.03.2005 |
| 10029626 | WINGLET TECHNOLOGY LLC | P-EASA.IM.A.S.02964 | | | CESSNA 750 | Installation of Elliptical Winglets on Cessna 750 aircraft/Removal of existing wingtips and installation of elliptical winglets,structural reinforcement of the wings and LED position & anti-collisionlights. | EASA TCDS IM.A.0977The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/ approvalCS 25.445 Amendment 1CS 25.1316 Amendment 1CRI F-301: External Lights (...) | Master Drawing List D10-2000 Revision B from 16 June 2009Airplane Flight Manual Supplements (AFMS):-AFMS 75EUA-S1WT-00 Revision 0 from 29 March 2010 for Cessna 750 S/N750-0173 andon and airplanes incorporating SB 750-71-10-AFMS 75FMA-S39-00 Revision (...) | None | 08.04.2010 |
| 10015353 | WIPAIRE Inc. | EASA.IM.A.S.00601 | REV. 4 | FAA STC SA01795CH | AT-802, AT-802A | Installation of Wipaire Model 10000 Amphibious floats per Wipaire Inc.STC SA01795CH, FAA approved July 8, 2011. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | FAA STC No.SA01795CH amended July 8, 2011 For Model AT-802A: Installation of Wipaire Model 10000 Amphibious Floatsin accordance with Wipaire, Inc. Installation Drawing No. 7D1-4251, NoRevision, dated January 13, 2003. Operations with the floats remo (...) | FAA STC No. SA01795CH "Limitations and Conditions", amendedJuly 8,2011.1. FAA approved Airplane Flight Manual Supplements:AT-802A amphibian, dated March 13, 2003AT-802A landplane, dated October 7, 2003AT-802 amphibian, dated September 29,2008AT-802 (...) | 31.08.2011 |
| 10015657 | WIPAIRE Inc. | EASA.IM.A.S.01633 | | FAA STC SA00674CH | CESSNA 172 SERIES | Installation of Wipaire Model 2100 Amphibian/Seaplane floats on Cessna/Reims Aviation 172/F172 Models as defined in theAML. | | Installation on accordance with Wipaire Inc Installation Drawings as listed on the AML to this STC, at latest approved revision.The appropriate FAA approved Airplane Flight Manual Supplement required for all models with floats installed is contained (...) | 1. FAA Supplemental Type Certificate SA332GL is required for models 172L, 172J, 172K and 172L when installed in accordance with Wipaire Drawing 10D1-6012.2. FAA Supplemental Type Certificate SA332GL is optional for models 172M, 172N and 172P when in (...) | 21.06.2006 |
| 10015832 | WIPAIRE Inc. | EASA.IM.A.S.01923 | | FAA STC SA2CH | BOMBARDIER DHC-6 SERIES 300 | Installation of Wipaire Model 13000 Amphibious Landing Gear. | | FAA STC SA2CHFAA Approved AFM Supplement for Amphibious Operation at Revision 5 dated 18 September 2007 or later approved revision.FAA Approved AFM Supplement for Seaplane Operations at Revision C dated 21 September 2007 or later approved revision. | Restricted to Amphibious or Seaplane Operation for Aircraftwith PT6-27 Engines only. | 10.10.2007 |
| 10015833 | WIPAIRE Inc. | EASA.IM.A.S.01924 | REV. 2 | FAA STC SA39CH | SEE EASA APPROVED MODEL LIST | Installation of an Amphibian Landing Gear Position AdvisorySystem | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertificated noiseand/ or emissions levels of the original product areunchanged and remain applicable to this certificate/ approval. | Lake & Air Installation Manual Report Number 9600-1A revision number anddate as listed on the EASA Approved Model List;AFM Supplement revision number and date as listed on the EASA ApprovedModel List, or later revisions of the above listed documents (...) | Prior to installation of this modification it must be determined thatthe interrelationship between this modification and any other previouslyinstalled modification and/ or repair will introduce no adverse effectupon the airworthiness of the product. (...) | 22.08.2013 |
| 10015891 | WIPAIRE Inc. | EASA.IM.A.S.02011 | | FAA STC SA610GL | | Installation of Wipaire Model 6000 or 6100 Seaplane or Amphibious Floats | | In accordance with Wipaire Master Drawing List ref W-MD115,dated, June 3 1982, or later approved revisions.FAA Approved Flight Manual Supplement dated, June 3, 1982 is required for Amphibious float installations, or later approved revisions.FAA Appr (...) | | 20.06.2006 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|------------------------|----------------|----------------------|--|---|---|---|--|---------------|
| 10028790 | WIPAIRE Inc. | | | FAA STC SA01397NY | 208, 208B | Installation of ERDA, Inc. Executive Seat, Model 3058 and folding table in acc. with FAA STC SA01397NY. | As defined in FAA TCDS A37CEThe Certification Basis for the original product remains applicable to this certificate/ approvalThe certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | -Wipaire, Inc. Document Control List SU8003DCL, Rev. C, dated November 5, 2009.-FAA approved Wipaire, Inc. Flight Manual Supplement SU8003FMS, Rev. C, dated November 24, 2009.-FAA accepted Wipaire, Inc. Instructions for Continued Airworthiness SU8003ICA (...) | None | 08.02.2010 |
| 10029642 | WIPAIRE Inc. | | | FAA STC SA914NE | CESSNA 206 SERIES | Installation of Wipaire, Inc. wing tip extensions in accordance to FAA STC SA914NE. | The Certification Basis for the original product remains applicable to this certificate/ approval.The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Wipaire, Inc. Master Drawing List number W-MD126, dated April 30, 2001 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | Placard "VNE = 165 Knots" is required with this installation. | 08.04.2010 |
| 10029679 | WIPAIRE Inc. | | | FAA STC SA1470GL | 1) CESSNA 206H, T206H 2) CESSNA 206, P206, P206A, P206B, P206D, P206E, U206, TP206C, TP206D, TU206C, TU206D TU206E, TU206F, TU206G U206A, U206B, U206C, U206D, U206E, U206F, U206G, TP206A, | Installation of right front cabin door in accordance with Wipaire, Inc. installation drawing number 3D1-1101 | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Installation drawing number 3D1-1101, no revision, dated February 15, 1990 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) | None | 12.04.2010 |
| 10030272 | WIPAIRE Inc. | | | FAA STC SA00059W1 | CESSNA 208, 208B | Installation of the Wipaire Single Point Fueling System on Cessna 208i.a. FAA STC SA00059W1 | The Certification Basis for the original product remains applicable to this certificate/ approvalThe requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval | Airplane Flight Manual Supplement, Doc.-No. 1004430, Rev. A, dated March 21, 2010.- Master Drawing List & Parts Manufacturer Approval List W-MD137, Rev. B, reissued February 19, 2010.-Wipaire, Inc. Installation Manual, Doc.-No. 1004433, Rev. A, dated Feb (...) | None | 08.06.2010 |
| 10031133 | WIPAIRE Inc. | | | FAA STC SA02808CH | DHC-6-200, -300 | Installation of nine passenger executive seating system in accordance with FAA STC SA02808CH | The Certification Basis for the original product remains applicable to this certificate/ approvala) The type certification basis for the DHC-6-200 and DHC-6-300 is as shown on TCCA TCDS A-82 for parts not changed or not affected by the change.b) The ce (...) | FAA approved Flight Manual Supplement, Document Number W056144-25, no revision, dated February 8, 2010. Instructions for Continued Airworthiness, Document Number W056144-24, no revision, dated February 11, 2010. Installation in accordance with Wipair, Inc. (...) | none | 28.07.2010 |
| 10032352 | WIPAIRE Inc. | P-EASA, I.M.A.S. 01058 | | FAA STC SA01916CH | AIR TRACTOR AT-802, AT-802A | Installation of vortex generators. Installation of Wipaire stall improvement array in accordance with Wipaire installation drawing No. 7A-1-4407 Revision B, dated April 27, 2010 for the AT-802A model and installation drawing No. 1004245, Revision B, dated April 27, 2010 for the AT-802 model | The Certification Basis for the original product remains applicable to this certificate/ approval | FAA STC No. SA01916CH amended May 24, 2010 "Installation of Wipaire Stall Improvement Array" for AT-802 and AT-802A models. W-AT802-67F-1424-VG. FAA approved AFMS for the AT-802 with PT6A-67F(1424shp) engine and Wipaire stall improvement array dated Ma (...) | FAA STC No. SA01916CH "Limitations and conditions", amended May 24, 2010 | 05.11.2010 |
| 10041960 | WIPAIRE Inc. | | | FAA STC SA1311GL | CESSNA 208 | Modification of Cessna 208 in accordance with FAA STC SA1311GL.1) Installation of Wipaire Model 8000 Amphibian/Seaplane floats or 2) Installation of Wipaire Model 8750 Amphibian/Seaplane floats and installation of vortex generators law. Wipaire STC SA2 (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certificated noise and/or emissions levels. | Ref. to 1): Wipaire Drawing No. 6C1-2023, Rev. C, dated March 20, 2012 (8000 lbs. gross weight) (Ref. to 2): Wipaire Drawing No. 1006000, Rev. A, dated June 5, 2012 (8750 lbs. gross weight) Pilot's Operating Handbook Supplements: Airplane serial No. ..., Mo (...) | 1. Airplanes not incorporating STC SA270CH, with Model 8000 floats: Airspeed Limits., VA at 8000 lbs. = 143 KIAS Maximum Weight., 8000 lbs. , Take off and Flight., 7800 lbs. , Landing (Water and Land), 8035 lbs. , Ramp., C.G. Range., Take off and Flight, (...) | 26.10.2012 |
| 10042954 | WIPAIRE Inc. | EASA, I.M.A.S. 01140 | | FAA STC SA00900CH | SEE EASA APPROVED MODEL LIST | Installation of Wipaire Model 2350 Amphibian/Seaplane floats on Cessna 172 Models (validation of FAA STC SA00900CH). | | Installation in accordance with Wipaire Inc. MDL W-MD100 issue date 25 July, 2005 or later approved revisions. | 1. Installation on Cessna models 172R and 172S require aircraft fitted with McCauley Model 1A200/WFA propeller and Textron Lycoming Model IO-360-L2A engine rated at 180hp and 2700 rpm. 2. FAA STC SA332GL is required for models 172i, K, L, M, N and P (Note: (...) | 04.07.2005 |
| 10015702 | WORLDWIDE AIRCRAFT SERVICES Inc. | EASA, I.M.A.S. 0169 | REV. 1 | | AEROSPATIALE ATR 42-200 AEROSPATIALE ATR 42-300 AEROSPATIALE ATR 42-320 AEROSPATIALE ATR 42-500 AEROSPATIALE ATR 72-101 AEROSPATIALE ATR 72-102 AEROSPATIALE ATR 72-201 AEROSPATIALE ATR 72-202 AEROSPATIALE ATR 72-211 AEROSPATIALE ATR 72-212 AEROSPATIALE ATR 72-212A | Passenger to all cargo conversion of ATR-42/72 series aircraft. ATR-42 is modified in accordance with Worldwide Aircraft Services Master Drawing List DL42-25-001 Rev G dated JAN 22, 2004. ATR-72 is modified in accordance with Worldwide Aircraft Serv (...) | | AFMS ATR 42-LC dated JAN 23, 2004 for ATR 42 aircraft and AFMS 72-00001-017 Revision IR dated SEP 2, 2005 for ATR 72 aircraft. ICADW Document Ref 72-00001-003 Revision A dated FEB 14, 2005 for ATR 72 | | 24.11.2005 |
| 10015808 | WORLDWIDE AIRCRAFT SERVICES Inc. | EASA, I.M.A.S. 01877 | | FAA STC ST00798W1 | FAIRCHILD SWEARINGEN SA 227-AC FAIRCHILD SWEARINGEN SA 227-AT FAIRCHILD SWEARINGEN SA 227-BC FAIRCHILD SWEARINGEN SA 227-CC FAIRCHILD SWEARINGEN SA 227-DC | Installation of Fairchild Digital Flight Data Recorder F800r F1000 in Fairchild aircraft SA227-AT, -AC, BC, CC, DC in accordance with Worldwide Aircraft Services Certification Drawing list No. DL227-31-06 Revision A, dated May 14, 1999 or later approved revision. | | Worldwide Aircraft Services Certification Drawing List No. DL227-31-06 Revision A, dated May 14, 1999 or later approved revision. FAA approved worldwide Aircraft Services Airplane Flight Manual Supplement, dated June 8, 1999, or later approved revision. | The FAA approved Airplane Flight Manual Supplement must be on board the aircraft at all times. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility wt (...) | 02.11.2007 |



| EASA STC no. | STC holder | Old STC ref. | Revision Level | Foreign/NAA STC ref. | Model | Description | Certification Basis | Limitations | Conditions | Date of issue |
|--------------|----------------------------------|----------------|----------------|----------------------|--------------------------------|--|---|---|---|---------------|
| 10031901 | WORLDWIDE AIRCRAFT SERVICES Inc. | | | FAA STC SA4872SW | SA226-T, SA226-AT | Installation of 2 Garrett AiResearch engines Model TPE 331-10UA-501G or TPE 331-10UA-511G | The Certification Basis for the original product remains applicable to this certificate/ approval. CAR 3 This certificate/ approval involves a change to the requirements for environmental protection and/or a change to the certificated noise and/or emissions levels. | SA226-T: Installed law KMA Information and Instructions No 401AD01, Rev EFAA approved AFM-S, dated 20 Dec 1982 SA 226-AT: Installed law KMA Information and Instructions No 401BD01, Rev EFAA approved AFM-S, dated 15 Apr 1985 or later revisions of the abo (...) | according limitation section of AFM-S | 21.09.2010 |
| 10032538 | ZF LUFTFAHRTTECHNIK GmbH | | REV. 2 | | MBB-BK 117 C-2 | Main Gear Box Eurocopter BK117 C2 Conversion of the Main Gear Box from MBB-BK117 C-1 built standard Part Number (P/N) 1171200501 to MBB-BK117 C-2 built standard P/NB632K1001-051 within the scope of a major overhaul, as documented in the ZF Luftfahrttec (...) | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Airworthiness Limitations:- ZFL doc. no. STC03-1682-1109, "Supplement for ECD Master Servicing Manual MBB-BK117 C-2, Chapter 04-00-00, 2-1", Issue 2 or later revisions of the above listed documents approved by EASA- Instructions for Continued Airworth (...) | - This conversion can only be performed by the STC holder.- After conversion to Main Gear Box P/N B632K1001-051 this is only to be installed on MBB-BK117 C-2 helicopters. | 13.06.2012 |
| 10031018 | ZIM FLUGSITZ GmbH | | | | A310-300 | Replacement of E/C seats by ZIM E/C seats. Type ECO-01 This STC consists in replacing the existing seats by Type ECO-01 seat. | The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Declaration of Compliance, ref : 502101-COC-003 Issue 0A, dated 27 July 2010 or later revisions of the above listed documents approved by EASA | None | 21.07.2010 |
| 10042845 | ZIM FLUGSITZ GmbH | | REV. 1 | | B737-800 | Installation of ZIM Flugstz ECO-01 EC Seats in a B737-800 Cabin | The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:- CRI F-01 on IFEdated 19 December 2012 The certificated noise and/or emissions levels o (...) | Master Document List (MDL), Reference 505402-MDL-002, Revision 0D, dated 19 December 2012- Airplane Flight Manual Supplement (AFMS), Reference 505402-AFM-001, Revision 0B, dated 17 December 2012- Weight and Balance Manual, Reference 505402-WBM-001, (...) | 1) The In-Seat IFE system shall not be used during Taxi, Take-Off and Landing. 2) The ZIM Flugstz ECO-01 EC Seats installation in accordance with this STC is limited to the following B737-800 aircraft S/N 41047 and 41190. | 21.12.2012 |
| 10013965 | ZOREX S.A | EASA.A.S.00659 | REV. 1 | | FAIRCHILD SWEARINGEN SA 226 TC | COMMNAV/GPS Garmin GNS430 Installation, for B-RNAV Operation, and COMM Bendix King KY196A Installation, both for 8.33 kHz standard. | | Doc "Programa de Certificación. Instalación Equipo COMMNAV/GPS Garmin GNS430, e instalación Equipo Comunicaciones KY196A", Edición 1. Installation Instructions in accordance with:- Anexo 3 al programa de certificación "Descripción del Montaje", (...) | 1. The aircraft is approved for dual pilot B-RNAV operation, subjected to the conditions, limitations and procedures contained in:- Anexo 8 al Programa de Certificación "Aircraft Flight Manual Supplement", Ed. inicial. 2. This STC is approved on (...) | 25.04.2006 |