

FINITE ELEMENT DYNAMIC SHAKEDOWN
ANALYSIS OF A GRAVITY TYPE OFFSHORE
STRUCTURE-FOUNDATION SYSTEM

CENTRE FOR NEWFOUNDLAND STUDIES

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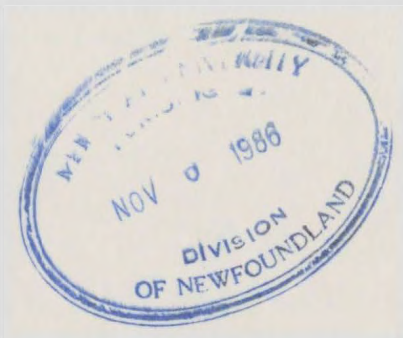
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VOLUME I

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FINITE ELEMENT DYNAMIC SHAKEDOWN
ANALYSIS OF A GRAVITY TYPE OFFSHORE
STRUCTURE-FOUNDATION SYSTEM

by

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ABSTRACT

The shakedown deformation of a gravity type offshore foundation subjected to dynamic cyclic wave loading is determined based on the two dimensional finite element formulation of a fluid saturated porous medium. The soil is assumed to behave in an undrained manner and modelled as a two-phase material. The undrained characteristics of the soil leads the Poisson's ratio to be one-half and thus results in a singular stiffness matrix. The reduced integration technique is used in order to circumvent the problem. The dynamic shakedown load factor is obtained by first computing the elastodynamic response of the caisson-foundation (saturated) medium for some adjusted initial conditions (e.g. displacement and velocity) and then solving a linear programming problem based on the dual form of kinematic shakedown theory.

A finite element computer code OPFA (Offshore Platform Foundation Analysis) is developed which computes the dynamic shakedown load factor of a gravity type offshore foundation and then determines the response quantities such as displacements and stresses in the saturated soil medium. The nonlinear stress-strain characteristics of the soil medium is considered in the analysis by use of an equivalent linear and an elastic-perfectly plastic model.

Several problems are solved in order to check the accuracy of the computer programme. The results are in excellent agreement with the available analytical solutions. The shakedown analysis for a flexible foundation shows that limit pressures depend on the inclination angle and the eccentricity of the load. The shakedown limit pressure decreases as the eccentricity and the angle of inclination of the load increase. Also, shakedown analyses indicate that the footing pressures are below those predicted by the approximate bearing capacity formula. These shakedown pressures are obtained from an upper bound solution and therefore the use of the above formula will not produce a conservative estimate of the bearing capacity of the foundation when the loading is cyclic in nature.

The dynamic shakedown load factor for the gravity foundation is only 10% lower than its static counterpart. The reason for the decrease of this load factor is due to the amplification of stresses under dynamic loading condition. In computing the response quantities for three different soil models, it is observed that the equivalent linear analysis predicts lower deformation as opposed to other linear and nonlinear models. However, for the quasi-static analysis of the gravity foundation, the permanent vertical shakedown deformation is within 0.085% of the footing width. In all cases it is observed that shakedown occurs with respect to the deformation of the foundation.

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LIST OF SYMBOLS

ENGLISH LETTERS

t = time parameter

\bar{x} = cartesian coordinate system (e.g. x and y)

p =, excess pore pressure over hydrostatic

n = porosity

k = permeability

X_B and Y_B = body force of bulk solid in x and y directions

X_F and Y_F = body force of fluid in x and y directions

u and w = displacement of solid and fluid with respect to solid

K_w = bulk modulus of water

M_o = modified bulk modulus of water

$\hat{\quad}$ = prescribed value

V = volume

S = total surface area

$\hat{T}(\bar{x}, t)$ = prescribed traction

$\hat{p}(\bar{x}, t)$ = prescribed fluid pressure

LIST OF SYMBOLS (Cont'd.)

l and m = direction cosines in x and y

h_i = i-th interpolation function

r and s = local coordinate system

x and y = global coordinate system

\underline{U}_x and \underline{U}_y = nodal solid displacement vectors
in x and y directions

$\underline{W}_{\approx x}$ and $\underline{W}_{\approx y}$ = nodal fluid displacement vectors
in x and y directions

\underline{U} and \underline{W} = displacement vectors for solid
and fluid at nodes

D_{ss} , D_{sf} and D_{ff} = stress-strain matrix for solid,
solid-fluid coupling and fluid
respectively

\underline{D} = matrix containing D_{ss} , D_{sf} and D_{ff}

M_{ss} , M_{sf} and M_{ff} = solid, solid-fluid and fluid mass
matrices for an element

C_{ss} and C_{ff} = solid and fluid damping matrices
for an element

K_{ss} , K_{sf} and K_{ff} = solid, solid-fluid and fluid
stiffness matrices for an element

D_{sd} = drained soil skeleton
constitutive matrix

p_B and p_f = bulk solid and fluid load vectors
for an element

P_B = assembled global load vector

LIST OF SYMBOLS (Cont'd.)

K_c = modified bulk modulus

E_u and E_d = undrained and drained Young's moduli respectively

D^s and D^v = deviatoric and volumetric constitutive matrices

S^e = strain energy

K_1 and K_2 = drained stiffness matrix and stiffness matrix due to volumetric strain respectively

B_1 and B_2 = Rayleigh damping coefficients

s_u = undrained shear strength

f_y = yield function

\hat{T}_i = prescribed surface traction

B_i = body force

u_i and \dot{u}_i = displacement and velocity field

$Q(.)$ = plastic dissipation density function

N = total number of cycles

\hat{C}_{kl} = coefficients of elasticity matrix

$F(\hat{T}_i)$ and $F_o(\hat{T}_i)$ = arbitrary loading and unit loading domain respectively

LIST OF SYMBOLS (Cont'd.)

$\bar{T}_s(\bar{x})$ = surface traction in s-th load mode

\tilde{N}_i = stress gradient matrix containing unit normal vector for i-th yield plane

f_{yi} = linearized i-th yield plane

H = hardening parameter

K_{oi} = vector containing the distance of the i-th yield plane from the origin

\tilde{N}_i^j = unit normal vector related to i-th yield plane of j-th element

R = total number of yield planes

V_j = subvolume for element, j

M = total number of elements

N^0 = total number of degrees of freedom (D.O.F.)

W_{int} = internal work

$[G_0]$ = global compatibility matrix

W_{ext} = external work

$\underline{U}(\bar{x}, t^*)$ and $\dot{\underline{U}}(\bar{x}, t^*)$ = fictitious displacement and velocity

t^* = arbitrary fictitious time

LIST OF SYMBOLS (Cont'd.)

$U(\bar{x}, 0)$ and $\dot{U}(\bar{x}, 0)$ = actual initial conditions;
displacement and velocity

\tilde{M}_i^j = maximum stress vector of j -th
element with respect to i -th
yield plane

\bar{U}^* and $\dot{\bar{U}}^*$ = displacement and velocity
computed at a finite time

T_D = dead load

G = secant shear modulus

G_{\max} = maximum shear modulus

D_{\max} = maximum damping ratio

m_j , c_j , \bar{K}_j = mass, damping and stiffness
matrices for element, j

D_j = damping ratio for element, j

M^* , C^* and K^* = assembled mass, damping and
stiffness matrices respectively

a and b = material parameters for soil

Y = hardening parameter

$[D_{sd}^{Ep}]$ = undrained elastic-plastic
constitutive matrix

c' = cohesion

J_1 and J_2 = first and second stress invariant

S_x , S_y and S_z = deviatoric stress components

LIST OF SYMBOLS (Cont'd.)

\bar{e}^p = effective plastic strain

\tilde{R} = residual load vector

ΔU = incremental displacement vector

$\hat{K}_t^{(i)}$ = effective undrained stiffness matrix for time t , at a particular iteration, i

\hat{P}_B = effective load vector

$\|\cdot\|$ = norm

RTOL = specified tolerance value

Δt = incremental time step

F_V = submerged or dead weight of the structure

q_{ult} = ultimate bearing pressure

N_c = bearing capacity factor = $(2+\Pi)$

e = eccentricity of the loading

B and A = footing width and area of footing

B' = effective width of the footing
= $B(1-2e/B)$

F_H and \bar{F}_H = horizontal loading and prescribed value

Q' = inclined loading

a_0 = height of the caisson structure

LIST OF SYMBOLS (Cont'd.)

F_M and \bar{F}_M = overturning moment caused by the horizontal component of the wave loading and its prescribed value

t_s = stiffness ratio of the caisson structure to foundation soil

GREEK LETTERS

σ_x , σ_y and τ_{xy} = total stress components

ρ and ρ_f = solid and fluid weight densities

ϵ_x , ϵ_y and γ_{xy} = strain components

ξ = volumetric strain of fluid with respect to solid

α = grain compressibility

λ and μ = Lamé's constants

λ_c = modified Lamé constant

ϕ' and ψ' = arbitrary functions

δW = virtual work

σ and ϵ = stress and strain vectors

$\underline{\phi}_m$ and $\underline{\phi}_m$ = matrix containing interpolation functions

$\underline{0}$ = null vector

LIST OF SYMBOLS (Cont'd.)

$\underline{\psi}$ = strain displacement matrix

ψ_{11} = solid strain-displacement matrix

ψ_{22} = fluid volumetric strain
displacement matrix

$\delta \underline{U}$ and $\delta \underline{W}$ = virtual displacement of solid and
fluid with respect to solid
respectively

ϵ_v = volumetric strain

ν_u and ν_d = undrained and drained Poisson's
ratios respectively

ϕ_u = angle of internal friction

σ_{ij}^s = safe soil stress tensor

σ_{ij}^a = admissible soil stress tensor

ϵ_{ij}^p and $\dot{\epsilon}_{ij}^p$ = plastic strain tensor and its rate

σ_{ij} = elastic plastic stress

σ_{ij}^E = elastic stress tensor

σ_{ij}^R = residual stress tensor

τ = time interval

β = load factor

β_s = shakedown limit load

LIST OF SYMBOLS (Cont'd.)

β_E = elastic limit load

ω and ω^* = period and frequency ratio

$\Delta\epsilon_{ij}$ = incremental strain tensor

$\Delta\epsilon_{ij}^E$ and $\Delta\epsilon_{ij}^P$ = incremental strain tensor;
elastic and plastic respectively

γ_s = s-th load parameter

σ_{ij}^{kE} = elastic stress field corresponding to load parameter, k

σ^T = transpose of stress vector

Δ = plastic multiplier

σ^j and ϵ^j = stress and strain for j-th element

σ_{\max}^E = maximum elastic stress response vector

σ_{\max}^{*E} = fictitious maximum elastic stress response vector

τ_{\max} = maximum shear stress

γ_{\max} = maximum shear strain

σ_{ii} = hydrostatic stress

S_{ij} = deviatoric stress tensor

δ_{ij} = Kronecker delta

LIST OF SYMBOLS (Cont'd.)

σ'_{ij} and σ' = effective stress vectors

σ_y and $\bar{\sigma}$ = yield stress

θ = parameter normally equals to 1.4

δ_H and δ_V =, horizontal and vertical deformations of the footing

α_A = load inclination angle

Δ^j_{\sim} = plastic multiplier for an element, j

$\Delta_{\sim} =$ plastic multiplier for all elements in assembled form

σ_v = normal vertical pressure

CHAPTER I

INTRODUCTION

1.0 Nature of the Problem

The continuous demand for oil and gas has challenged ocean engineers to undertake the design and construction of large gravity type offshore structures in hostile ocean environments. The gravity type offshore structure has an obvious advantage over the steel jacket type platform because of its shorter installation time and ability to provide the oil storage facilities in its cellular compartments. The large magnitude of wave forces which are transmitted by the structure to the surrounding soil cannot be compared with a similar land based structure. As the platforms are being installed in deeper water, the complexity of the problem has increased enormously. Conventional methods of analyses are no longer applicable and new techniques need to be developed for complete dynamic analysis of the structure-foundation system.

A typical offshore gravity structure will always encounter two situations: a) calm water condition and b) storm condition. During the passage of

large waves, the net force on the structure will vary and reach its maximum value as the wave crest strikes the structure. Fig. 1.1 depicts the principal forces that will act upon the structure as the wave passes. It is seen from Fig. 1.1, that the structure is subjected to the cyclic nature of loading, which in turn will generate cyclic stresses in the foundation soil. The duration of this cyclic loading can be for several hours with a relatively low frequency range of about 0.05 - 0.5 Hz. The dynamic nature of the wave forces on the foundation causes great concern in the geotechnical design aspect and therefore requires the development of a reliable method for the analysis of offshore structure-foundation problems.

The soil behaviour beneath the foundation of an offshore structure subjected to dynamic cyclic loading may be separated into a short-term undrained condition with no drainage allowed and a long-term condition in which drainage occurs. In the undrained short term condition, the excess pore pressure leads to either a reduction or an increase of 'effective stress' with accompanying reduction or increase in soil stiffness and strength. For the platform this means an increase or decrease in the natural period of vibration. The magnitude of

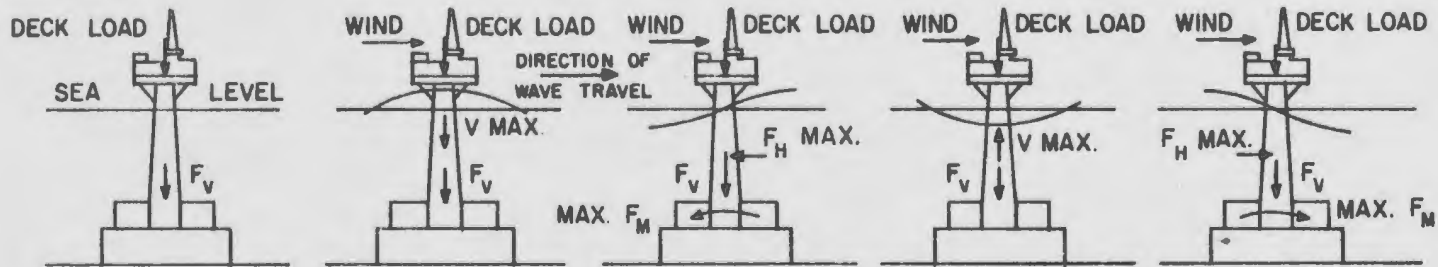
WAVE PHASE ANGLE

0 or 2π

$\pi/2$

π

$3\pi/2$



F_m = OVERTURNING MOMENT

APPROX. GROUND PRESSURE DISTRIBUTION



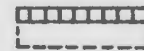
(a)



(b)



(c)



(d)



(e)

FIG. 1.1 CYCLIC LOADING ON THE FOUNDATION OF A GRAVITY TYPE OFFSHORE STRUCTURE DUE TO WAVE ACTION (STUBBS, 1976)

changes in 'effective stresses' due to the change in average pore pressure (increase/decrease) will depend on the rate of dissipation of pore water. It is seen from the above argument that the 'effective stress' is responsible for all major deformations, linear or nonlinear, and failure states can only be adequately modelled in terms of such 'effective stress'. It has been shown that the soil properties, in particular the stiffness, are among the most important factors influencing the dynamic behaviour of typical offshore gravity platforms.

For foundations subjected to loads varying in time in a nonproportional manner within prescribed limits, the classical limit theorems can give unsafe estimates of the collapse load, as failure can occur at loads well below the static collapse values. Shakedown theorems, which are generalizations of the limit theorems, can provide appropriate safe bounds for such complex loading programmes. For load reversal varying between prescribed limits, a foundation is said to shakedown when a state is reached such that all subsequent load applications produce only elastic changes. In recent years, very few attempts have been made to find practical solutions to the

shakedown problem related to marine foundations resting on saturated soils. As the nonlinear dynamic response analysis of an offshore foundation to wave loading can only be carried out for a very limited number of cycles due to constraints on computing costs, the shakedown analysis becomes more practical and relevant to the overall design process.

Research in the field of earthquake engineering has shown that the soil deformation can vary from small to very large amplitude depending on the nature of the dynamic loading. Cyclic loading due to wave causes permanent deformation and thus strain under both drained and undrained conditions. The permanent strains are cumulative in nature i.e. carried over from one cycle to another and remain even if the cyclic loading has been terminated.

In predicting the permanent deformation of an offshore foundation system, one has to incorporate the appropriate soil model in a finite element analysis. Current research in soil stress-strain relationship is progressing towards the development of a three dimensional soil model based on the principles of plasticity as well as linear or non-linear elasticity. Several sophisticated constitutive

relationships have been proposed in recent years although none of them completely describe the behaviour of real soil.

1.2 Object of the Research

The complete dynamic response analysis of an offshore gravity foundation-soil system with the evaluation of the shakedown limit load which will further enable prediction of the permanent deformation of the foundation is beyond the present state-of-the-art.

The object of this research is to develop an analytical method for the prediction of the permanent deformation of an offshore foundation with particular reference to soil shakedown. A dynamic finite element formulation of the shakedown problem is presented for a fluid saturated porous medium.

1.3 Organization of the Thesis

Chaper 2 reviews the literatures related to the present investigation. These are described under three sections: 1) fluid saturated porous medium 2) constitutive modelling of soil and 3) shakedown analysis related to structural mechanics and geomechanics.

Chapter 3 presents the general dynamic finite element formulation of a saturated soil medium in which the soil is modelled as a two-phase material. The undrained characteristics of the soil lead the Poisson's ratio to be one-half and thus results in a singular stiffness matrix. The reduced integration technique is used in order to circumvent the problem. The two types of problems e.g. a strip footing and a cantilever beam are chosen as test cases in order to check the computer programme based on the present formulation.

Chapter 4 presents the shakedown formulation of the offshore foundation problem under cyclic loading conditions. Basic background on shakedown related theorems is presented in the first part whereas the finite element formulation of the dynamic shakedown problem is described in the second part. Typical boundary value problems are solved in order to validate the especially developed computer code.

Chapter 5 describes the various types of soil models. The general formulations for the equivalent linearized soil model and the incremental nonlinear elastic-plastic model are presented. Finally the chapter concludes with the computation of the shakedown displacement of a

strip footing under repeated vertical loading condition.

Chapter 6 describes the series of engineering problems where the accumulation of permanent displacements are studied. Three types of analyses are carried out. The first analysis presents the elastic and shakedown limit loads of a flexible foundation and the effects of the load eccentricity and the angle of load inclination on these limit loads. The second analysis presents the permanent deformation of a gravity type offshore foundation considering the static soil shakedown. In the third analysis, the dynamic shakedown load factor is first determined and the responses are then computed for various types of soil models. In the second and third analyses, the effect of the caisson structure is taken into account by modelling it as a stiff footing.

Chapter 7 reviews the results of the research and recommends possible extensions of the present work.

Appendix A presents the proofs of the shakedown theorems. Appendix B summarizes the intermediate steps in deriving the elastic-plastic constitutive matrix based on the formulation presented in

Chapter 5. Appendices C and D describe the computer code OPFA (Offshore Platform Foundation Analysis) in a detailed manner together with the preparation of input data. Appendix E in Vol. II of this thesis presents the listing of the computer program 'OPFA'.

CHAPTER II

REVIEW OF LITERATURE

2.1 Introduction

This chapter reviews the literature related to the present investigation. For convenience, the reviews are divided in three sections. These are a) fluid saturated porous media b) constitutive modelling of soil medium and c) shakedown analysis with particular reference to structural mechanics and geomechanics.

2.2 Fluid-saturated Porous Media

A systematic development of the mechanics of deformation of fluid saturated (fully) porous solids was presented by Biot (1941, 1955, 1956a, 1956b, 1956c, 1962). The basic concept in the development of the theory was to treat the saturated soil as a two-phase media with a matrix of solid particles (the soil skeleton) and the water filling the space not occupied by the solids (pore fluid). Biot's theory was based on pure mechanical considerations. Later Green and Naghdi (1965) established the formal basis of the dynamic

theory of mixtures from which Green and Steel (1966) formulated a theory of poroelasticity. Dereshewicz and Skalak (1963) have shown the uniqueness in dynamic poroelasticity using Neumann's uniqueness theorem of classical elasticity to liquid-saturated porous solids within the framework of Biot's theory. Recently Firoozbakhsh (1976) presented a unified treatment of the mechanics of deformation of a nondissipative liquid-filled saturated porous elastic medium based on Mindlin's work of microstructure in linear elasticity. Crochet and Naghdi (1966) developed the theoretical basis for the constitutive theory for flow of fluid through an elastic solid.

Since 1960, finite element approximations in conjunction with appropriate variational principles have shown great potential for the solution of the initial boundary value problems in different branches of engineering. Sandhu and Wilson (1969) developed a finite element scheme for theoretical field equations of fluid flow in saturated porous elastic medium based on Biot's theory using Gurtin's (1964) variational principle. The constitutive equation used for the solid phase was based on an elastic, isotropic, linear material behaviour. Later, Ghaboussi and Wilson (1972) extended the

quasistatic formulation of Sandhu and Wilson (1969) to the dynamic problem.

Finite element formulation of the wave propagation problem using Biot's field equations for dynamical condition was carried out by Melian and Brebbia (1972). Biot's equations were reformulated following a Galerkin type formulation to discretize the governing differential equations of motion. Taga and Togashi (1975, 1977) studied the similar wave propagation problem based on the Hamilton's Principle. Applying the bimixture theory and the finite element method, free vibration and steady state responses were obtained. Ishihara (1970) furnished an approximate form of the wave equations based on Biot's theory and tried to correlate these wave equations with those normally obtained from the ordinary elastic theory.

A rigorous finite element formulation based on Gurtin's variational principle was made for the governing equations in Biot's consolidation theory by Yokoo, Yamagata and Nagoka (1971a, 1971b, 1971c). Their analyses included idealization of nonhomogeneous, anisotropic soil as a one, two or three dimensional consolidation problem.

2.3 Constitutive Modelling

Cyclic loadings due to waves induce permanent strains in the soil foundation system. The permanent strains often termed as cumulative or residual strains will remain even if the loading has been terminated. Accumulation of permanent strains in the soil due to repetitive nature of loading such as wave or earthquake will depend primarily on the soil type and density, permeability, overconsolidation ratio, initial stresses, and finally number of frequencies and magnitudes of the loading cycles.

The current state-of-the-art on computational methods for permanent displacements of offshore gravity platform can be categorized by the following two approaches.

In the first approach an incremental analysis is carried out for the entire loading cycles which involves consideration of large number of time steps. Prevost (1977) has illustrated the use of a complex elasto-plastic soil model in order to predict the quasi-static response of an offshore foundation to wave loading. Prevost's model was based on the anisotropic theory of plasticity and has

often been criticized for using large number of material parameters. Prevost introduced a set of nested yield surfaces and the associated plastic moduli to model the undrained characteristics of soil such as clay. Mroz, Norris and Zienkiewicz (1978) have also presented a soil model similar to Prevost's model based on two surface anisotropic plasticity theory. The basic concepts behind the Mroz's theory are: a) the consolidation history is bounded and defined by Roscoe-Burland (1968) surface b) yield surface defines an elastic domain within the bounding surface and c) the plastic modulus varies from a value H_0 near the yield surface to a value H_b on the bounding surface. The Mroz's model is often considered as extension of the work of Dafalias and Popov (1976) who introduced the concept of vanishing 'yield surface' in modelling the soil behavior based on plasticity theory. Finn and Martin (1980) evaluated the merits of the above two models e.g. Prevost's model and Mroz's model and concluded that anisotropic plasticity model predicts the static response of soil in a reasonable manner. However, the verification of the model to cyclic loading problem was very limited at the time of their study.

Van Bekelen (1980) stated that the anisotropic plasticity model of soil can only produce some features associated with cyclic loading and that the verification of the model with respect to laboratory soil tests is very limited. He illustrated that the Mroz's model (1978) will predict a pore pressure under undrained loading which is too small for normally consolidated clays and will show the incorrect sign for heavily overconsolidated soil.

It is seen from the above review, that there is no unified soil constitutive modelling techniques available at this time to study the effects of load reversal under two and three dimensional stress conditions.

The second approach for predicting the permanent deformation of foundation under repetitive nature of loading has been described by several authors e.g. Anderson et al (1978), Dumas and Lee (1980), Marr and Christian (1981).

Dumas and Lee (1980) presented a step by step approach for the computation of permanent deformation of foundation to a small parcel of uniform cycles of wave loading. The loading was

treated as a linear static one and the following steps describe the procedure:

1. First the initial displacements, static stresses and cyclic stresses are computed.
2. From the laboratory test data, the strains are computed in each element based on the magnitude of stress level and number of cycles.
3. A pseudo-plastic modulus, E_p , is defined as the ratio between the cyclic stress, τ_e , to the strain, γ_e , computed in step 2.
4. Combined secant modulus, E_{ip} , is determined based on the initial modulus, E_i , and the plastic modulus, E_p , as $1/E_i + 1/E_p$.
5. Permanent displacements are then computed as the difference between the nodal displacements computed using secant modulus, E_{ip} , and the initial displacements based on tangent modulus, E_i .

Anderson (1978) first presented a graphical method based on the cyclic stress-strain data on saturated clay under undrained cyclic loading. The stress strain behaviour of a clay element subjected to two way cyclic loading in a simple shear apparatus under undrained condition is shown in Fig. 2.1. During the test, the cyclic shear stress, τ_{hc} , was kept constant (stress controlled test) at a

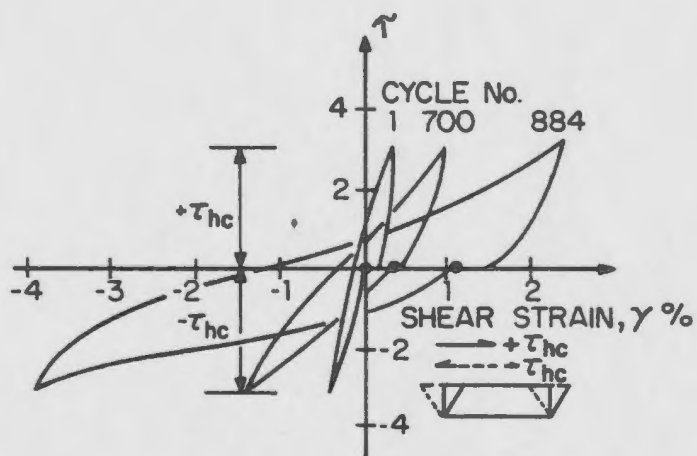


FIG. 2.1 STRESS-STRAIN CURVES FROM SIMPLE SHEAR TESTS
WITH UNDRAINED TWO-WAY CYCLIC LOADING
(ANDERSON, 1976)

level of $\pm 14.2 \text{ KN/m}^2$ which corresponds to 45% of the horizontal static shear stress, τ_{hf} , at failure under undrained condition. It can be seen from the Fig. 2.1, that the cyclic shear strain increases from +0.35% (0.0035) in the first cycle to +3.1% (.031) after 884 cycles. Based on these tests, a strain contour diagram, Fig. 2.2, is first constructed which shows the variation of cyclic shear strain with a given stress level and number of cycles. While using this method, waves are arranged in blocks, each representing wave loads of approximately equal amplitude. Each block of waves gives two contributions to the cyclic shear strain level. These are a) immediate cyclic strain level caused by the increase in stress level from one block to the next and b) increase in cyclic shear strain level due to the number of cycles at the particular stress level. The cyclic shear strain at the end of a block is defined as the initial value for the next block and is expressed as

$$\gamma_{C,N+\Delta N} = \gamma_{C,N} + \Delta\gamma_{C,i} + \Delta\gamma_{C,N} \quad \dots (2.1)$$

where $\gamma_{C,N+\Delta N}$ = cyclic shear strain after, $N+\Delta N$ cycles (Fig. 2.3)

$\gamma_{C,N}$ = cyclic shear strain in cycle N
with a cyclic shear stress, $\tau_{C,N}$

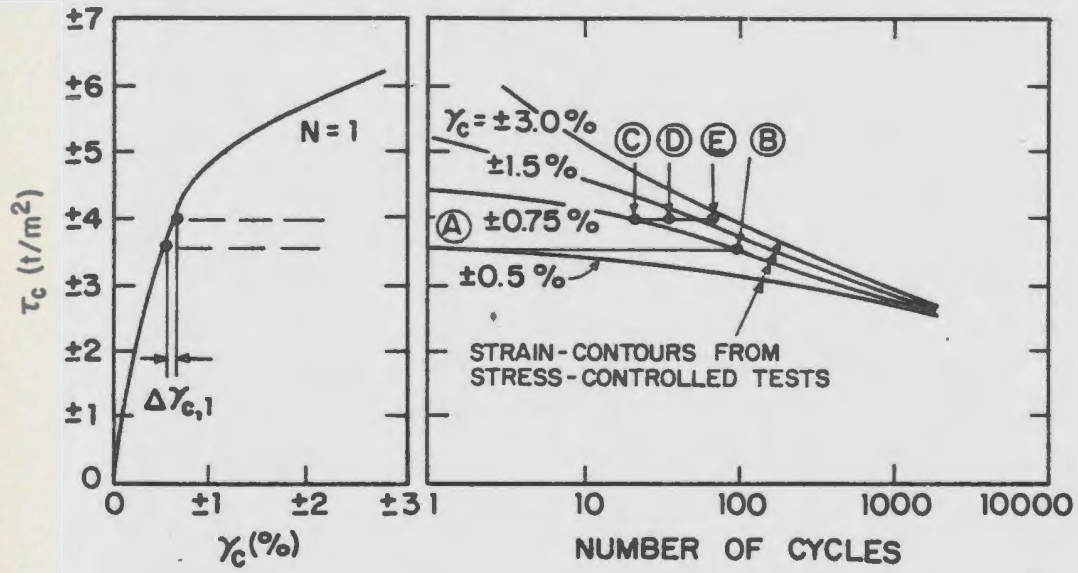


FIG. 2.2 PREDICTION OF CYCLIC SHEAR STRAINS FOR SOIL ELEMENT SUBJECTED TO VARYING CYCLIC SHEAR STRESSES. (ANDERSON, 1976)

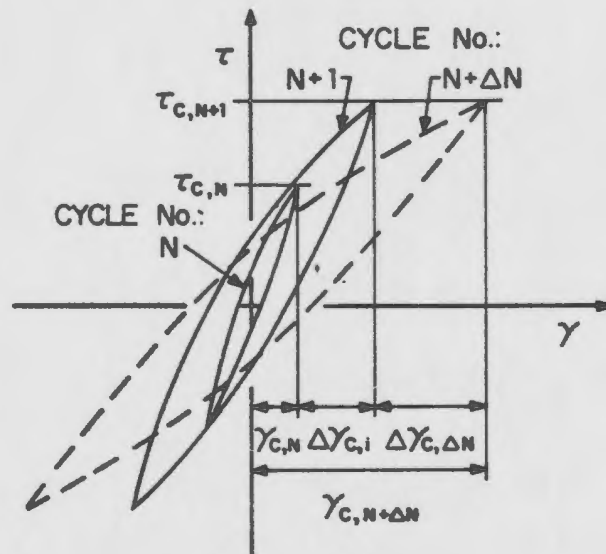


FIG. 2.3 STRESS-STRAIN BEHAVIOR OF SOIL ELEMENT
SUBJECTED TO VARYING CYCLIC SHEAR STRESSES
(ANDERSON, 1976)

$\Delta\gamma_{C,i}$ = immediate change in cyclic shear strain due to change in cyclic shear stress level from $\tau_{C,N}$ to $\tau_{C,N+1}$
 and $\Delta\gamma_{C,\Delta N}$ = increase in cyclic shear strain due to ΔN cycles with a cyclic shear stress, $\tau_{C,N+1}$

Marr and Christian (1981) described a method based on the concept of 'no net load'. The permanent strains for each element were evaluated during particular loading cycle and their effects represented by unbalanced equivalent nodal loads at the end of the cycle. Before the computation proceeds further, this unbalanced nodal load was redistributed resulting in additional deformations in the elements. The permanent deformations were then computed from the difference between the above two deformations. The constitutive relationship for soil was obtained based on a series of tests carried out on clay. They obtained a nearly log-linear relationship among cumulative strains, number of cycles, porosity, stress level, and cyclic shear stress.

2.4 Shakedown Analysis

2.4.1 Structural Mechanics

Bleich (1932) presented the static shakedown theorem with a limited proof. However, general proof of static shakedown theorem with the illustration of a truss problem was first presented by Melan (1936). Applications of static shakedown theory for continuous medium have been presented by Melan (1938), Symonds (1951) and Koiter (1960). Koiter (1960) also extended the work of Melan to the kinematic characterization of inadapation (non shakedown) and presented the basic proof for kinematic shakedown theorem. The extension of Melan's static shakedown theorem to temperature load cycles was formulated by Prager (1956).

Considerable work has been done on the application of shakedown theory for discrete structures (frames, trusses, etc.). Symonds (1951) was the first to present the application of shakedown theorem to a continuum problem. A circular shaft subjected to combined axial force and twisting moment was analysed and the shakedown domain for the axial force and twisting moment obtained. Leckie and Penny (1969) carried out a shakedown analysis for a complex structure based on Melan's theorem. They evaluated the lower bound estimates

of shakedown load factors for pressure, twist and moment applied to a radial flush of nozzle in a spherical pressure vessel. The residual stress field was approximated based on an available elastic solution and shakedown load factor finally evaluated using linear programming technique. Combining the loading and the temperature, Leckie and Penny (1969) showed how a shakedown analysis can help in making the important design decisions of a pressure vessel.

The matrix formulation of the shakedown problem was first presented by Maier (1969) with the use of piecewise linearization of the yield surface. Maier (1969) proposed that the shakedown load factor can be obtained as the optimal value of a linear programming problem. Melan's theorem was presented as a primal linear programming problem whereas Koiter's theorem was presented in its dual form. Koiter's theorem was also extended in order to allow for variable dislocations with associated flow laws. Maier (1969) also presented the shakedown theory based on nonassociated flow laws (non normality condition) and showed the procedures for obtaining the lower and upper bounds to the shakedown load factor.

Ceradini (1969, 1980) first presented the dynamic

shakedown theory as an extension to Melan's and Koiter's formulations. Dynamic shakedown theory is applicable to the situation where repeated loadings are varying so rapidly that the inertia and damping forces cannot be neglected (e.g. seismic-structure or wave-structure-foundation interaction problem).

The first dynamic shakedown theorem states that shakedown will occur if the yield condition is satisfied at every point and time by the sum of a constant residual stress field and the linear elastodynamic response evaluated on the basis of some arbitrary initial condition. Corradi and Maier (1972) extended the Koiter's inadaptation theorem to the dynamic problem and stated this as a second dynamic shakedown theorem. Ho (1972) formulated the dynamic shakedown theory of elastoplastic systems as an extension to Koiter's inadaptation theorem (second shakedown theorem). It was shown that shakedown design of a system under dynamic loading can be transformed into a quasi-static elastic counterpart with the proper application of dynamic amplification factor over its elastic response. Maier (1972) extended Ceradini's dynamic shakedown theorem to systems which obey general hardening rule, using the dual relationship between the static and kinematic

shakedown theorems; Koiter's theorem was also extended to allow for work hardening. Maier (1972) also included the effect of geometric nonlinearity in the classical shakedown theorem of Melan and Koiter and presented a proof for the new second order shakedown theory. König (1969) developed the shakedown theory of plates based on Melan's theorem using the yield locus concept; shakedown load factors were evaluated for simply supported and clamped plates by using the Tresca yield condition.

Although the shakedown theory based on finite element formulation has been developed for a broad range of problems, e.g. work hardening, dynamic and geometric nonlinearity, very few numerical solutions applicable to a continuum problem do exist in the literature. Belytschko (1972) presented the first numerical solution for shakedown analysis in continuum problem under plane stress condition. The problem was formulated with equilibrated finite element modelling coupled with a nonlinear programming approach. The method was applied to analyze the shakedown of a square plate with a circular hole subjected to biaxial variable repeated loadings. The material was assumed to be elastic-perfectly plastic obeying Von Mises yield condition. Corradi and Zavelani (1974) used the

approach of Maier, based on piecewise linearization of the yield surface coupled with compatible finite element displacement model, and analyzed the same problem as that solved by Belytschko (1972).

The same thin plate, plane stress problem was also analysed by Hung and Palgen (1979) based on the theory as formulated by Hung and König (1976). The formulation was a static one with the yield criterion defined as the 'yield criterion of the mean'. Borkowski and Kleiber (1980) presented the numerical shakedown analyses for a hyperstatic truss and for a cantilever beam under plane stress conditions. Kleiber, König, and Sawczuk (1982) presented an excellent report on studies related to plastic structures. Special topics which were covered in the above presentation were: stability, anisotropy, hardening and cyclic loads. Several example problems were illustrated showing the results regarding the computer analyses of non-linear structural mechanics problems.

2.4.2 Geomechanics

Very few applications of shakedown criteria for soil mechanics problems are available at present contrary to the structural mechanics problem as presented earlier.

Rowe (1975) carried out experimental work on model studies of an offshore gravity platform foundation resting on saturated clays. The foundation was subjected to inclined, eccentric, and cyclic quasi-static loads to simulate the wave loading on an actual platform. The 'amount of shakedown' was considered as a function of the size of the wave force and the number of wave cycles. The load factor was defined as the value corresponding to 'equilibrium shakedown' of the vertical displacements of the foundation.

Zienkiewicz (1976, 1978) presented the effect of wave action on the foundation of an offshore structure using finite element analysis. The primary conclusion was that even such a relatively slow cyclic loading may lead to 'progressive deformation' (incremental collapse) at load level well below the static collapse value. In the earlier publications of Zienkiewicz (1976, 1978), 'ratchetting' (progressive collapse) was defined as the shakedown phenomenon. In fact, this is contrary to the actual definition given by Bleich (1932), Melan (1936), Neal (1956) and later by Maier (1969).

Aboustitt and Reddy (1980) developed a computer program for the application of the kinematic static

shakedown theorem applicable to dry soil (drained) situation. The application was illustrated for a footing subjected to variable repeated loading under a plane strain condition. As there was no available solution for shakedown analysis in plane strain, the software developed was checked with the only available solution for plane stress: a square plate with a central circular hole, subjected to biaxial variable repeated loading. By making use of the fact that the shakedown theorems are generalizations of the limit theorems for a complex loading programme, the formulation was checked with available solutions of plane strain limit analysis. The footing analysis showed that the shakedown load varies almost linearly with the uniaxial compressive strength. At the same time Pande (1980) presented approximate solution of shakedown load for a strip footing under quasi-static loading situation. The traditional method of incremental nonlinear (elasto-plastic) analysis was used to obtain the lower bound on shakedown load factor. The soil was assumed to be purely cohesive and to obey the Mohr-Coulomb yield criterion. It was identified that shakedown analysis of offshore foundation problem will be of great interest to the geotechnical engineering field even though the clear procedures for carrying out such an analysis were lacking.

CHAPTER III

FINITE ELEMENT FORMULATION FOR FLUID
SATURATED POROUS MEDIUM3.1 Introduction

The general dynamic finite element formulation of saturated soil medium is described. The soil is modelled as a two-phase material whose state is described by the stresses and displacements in each phase. The undrained behaviour of fully saturated soil is accompanied by the condition of 'no volume' change and therefore the soil deforms like an incompressible material. The traditional displacement based finite element formulation does not apply when a material becomes truly incompressible. The incompressibility leads the Poisson's ratio to be one-half and thus results in a singular stiffness matrix. Reduced integration technique is used in order to circumvent the singularity problem.

The chapter is divided in three sections. The first section describes the general equations of motion of a fluid saturated porous medium based on Biot's approach. The second section presents the displacement based finite element formulation of the dynamic

equations of motion using reduced integration technique. Two types of loading, e.g. undrained and drained states of loading, are discussed with respect to soil foundation interaction. A computer code for the elastodynamic analysis is developed and its correctness checked by analysing two problems: a) a strip footing under vertical loading and b) a cantilever beam under periodic vertical loading.

3.2 Deformation Characteristics of Saturated Soil

The deformation analysis of saturated soil is normally carried out based on the assumption of the soil as a two-phase medium. The grains of the soil skeleton constitute the solid phase while the pore fluid filling the voids makes up the fluid phase. Normally, it is assumed that the soil skeleton behaves linearly or nonlinearly whereas the pore fluid is incompressible. The stresses in the soil skeleton are conventionally termed as "effective stresses" and those in the fluid (water) phase called the pore pressure.

During the loading process the two-phase characteristics of the saturated soil causes it to behave in two widely different ways. When the loading is applied in an instantaneous manner, excess pore

pressure develops and the pore fluid then tends to flow due to the difference in the pressure gradient within the medium. The flow then continues for a finite time depending on the permeability of the soil and the drainage boundary condition. At the instant of the loading the pore fluid has little time to dissipate and because the pore fluid is incompressible relative to the soil skeleton, the soil deforms as an incompressible material. The deformation of soil with 'zero volume change' is often defined as undrained deformation. When the soil is loaded with free drainage condition, such that excess pore pressures dissipate, the deformation is defined to be drained. However, in intermediate state where the flow can take place, interaction between the skeleton and the flow of the pore water must be considered. Terzaghi first observed the interaction in 1925, and he assumed that the water filling the pore volume behaves as an incompressible liquid and the deformation of the skeleton is elastic. The individual grain is considered as incompressible. Darcy's law is applicable due to the pore fluid flow being laminar. Terzaghi's one dimensional consolidation theory, however, is not applicable for most practical problems where the soil is loaded in multi directions. Biot (1941) presented an alternative approach based on multidimensional theory. Biot's theory is used in

the present study for the finite element formulation of the equations of motion for soil deformation analysis under dynamic loading. The basic assumptions in Biot's theory are:

1. Soil is modelled as a two-phase material.
2. The solid skeleton behaves in a linearly elastic form.
3. The fluid (water) is compressible and of the Newtonian type.
4. The individual grain constituting the soil skeleton also deforms elastically.
5. The total stress can be separated into two parts; a) effective stress and b) pore pressure.
6. Volume change due to shear deformation is not considered here.

3.3 Equations of Motion for a Saturated Porous Medium

The general equations of motion for fluid saturated porous medium were first presented by Biot and the

equations are for i) bulk fluid-solid mixture,

$$\frac{\partial \sigma_x}{\partial x} + \frac{\partial \tau_{xy}}{\partial y} + \rho b_x = \rho \ddot{u}_x + \rho_f \ddot{w}_x \quad \dots (3.1)$$

$$\frac{\partial \tau_{xy}}{\partial x} + \frac{\partial \sigma_y}{\partial y} + \rho b_y = \rho \ddot{u}_y + \rho_f \ddot{w}_y \quad \dots (3.2)$$

and ii) for the fluid with respect to the solid,

$$\frac{\partial p}{\partial x} + \rho_f b_x = \rho_f \ddot{u}_x + \frac{1}{n} \rho_f \ddot{w}_x + \frac{1}{k} \dot{w}_x \quad \dots (3.3)$$

$$\frac{\partial p}{\partial y} + \rho_f b_y = \rho_f \ddot{u}_y + \frac{1}{n} \rho_f \ddot{w}_y + \frac{1}{k} \dot{w}_y \quad \dots (3.4)$$

where σ_x , σ_y and τ_{xy} = total stresses of bulk solid
in x and y directions

$X_B = \rho b_x$ = body force of bulk solid in x
direction

$Y_B = \rho b_y$ = body force of bulk solid in y
direction

u_x , \dot{u}_x and \ddot{u}_x = displacement, velocity and
acceleration of bulk solid in
x direction

u_y, \dot{u}_y and \ddot{u}_y = displacement, velocity and acceleration of bulk solid in y direction

w_x, \dot{w}_x and \ddot{w}_x = displacement, velocity and acceleration of fluid relative to solid in x direction

w_y, \dot{w}_y and \ddot{w}_y = displacement, velocity and acceleration of fluid relative to solid in y direction

p = excess pore pressure over hydrostatic

$X_F = \rho_f b_x$ = fluid body force in x direction

$Y_F = \rho_f b_y$ = fluid body force in y direction

n = porosity

and k = permeability

3.4 Strain-displacement Relationship

The strain displacement relationships for the solid and the fluid are:

$$\epsilon_x = \frac{\partial u_x}{\partial x} ; \quad \dots (3.5a)$$

$$\epsilon_y = \frac{\partial u_y}{\partial y} ; \quad \dots (3.5b)$$

and

$$\gamma_{xy} = \frac{1}{2} \left(\frac{\partial u_x}{\partial y} + \frac{\partial u_y}{\partial x} \right) ; \quad \dots (3.5c)$$

The volumetric strain of fluid with respect to solid is given as

ξ = volumetric strain of fluid with respect to solid

$$= \frac{\partial w_x}{\partial x} + \frac{\partial w_y}{\partial y} \quad \dots (3.5d)$$

where ϵ_x , ϵ_y and γ_{xy} = solid strain components in x and y directions

and ξ = volumetric strain of fluid with respect to solid

3.5 Constitutive Relationships

Stress strain relationship for saturated porous media is expressed in terms of

$$\sigma_x = (\lambda_c + 2\mu) \frac{\partial u_x}{\partial x} + \lambda_c \frac{\partial u_y}{\partial y} + \alpha M_o \xi \quad \dots(3.6a)$$

$$\sigma_y = \lambda_c \frac{\partial u_x}{\partial x} + (\lambda_c + 2\mu) \frac{\partial u_y}{\partial y} + \alpha M_o \xi \quad \dots(3.6b)$$

$$\tau_{xy} = \mu \left(\frac{\partial u_x}{\partial y} + \frac{\partial u_y}{\partial x} \right) \quad \dots(3.6c)$$

$$p = \alpha M_o \frac{\partial u_x}{\partial x} + \alpha M_o \frac{\partial u_y}{\partial y} + M_o \xi \quad \dots(3.6d)$$

The above Eqns. (3.6a - 3.6d) can be written as

$$\begin{bmatrix} \sigma_x \\ \sigma_y \\ \tau_{xy} \\ p \end{bmatrix} = \begin{bmatrix} (\lambda_c + 2\mu) & \lambda_c & 0 & \alpha M_o \\ \lambda_c & (\lambda_c + 2\mu) & 0 & \alpha M_o \\ 0 & 0 & \mu & 0 \\ \alpha M_o & \alpha M_o & 0 & M_o \end{bmatrix} \begin{bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \\ \xi \end{bmatrix}$$

$$= \begin{bmatrix} \lambda + 2\mu & \lambda & 0 & \alpha M_o \\ \lambda & \lambda + 2\mu & 0 & \alpha M_o \\ 0 & 0 & \mu & 0 \\ \alpha M_o & \alpha M_o & 0 & M_o \end{bmatrix} \begin{bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \\ \xi \end{bmatrix} + \begin{bmatrix} \alpha^2 M_o & \alpha^2 M_o & 0 & 0 \\ \alpha^2 M_o & \alpha^2 M_o & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \\ \xi \end{bmatrix}$$

..(3.6e)

where $\lambda_c =$ modified Lamé's constant

$$= \lambda + \alpha^2 M_0$$

λ and $\mu =$ the Lamé's constants defining the properties of soil skeleton under drained condition

$\alpha =$ compressibility of grains
($n \leq \alpha \leq 1$)

$$M_0 = \frac{K_w}{n}$$

where $K_w =$ bulk modulus of water

3.6 Boundary Conditions

The boundary conditions are given as (Fig. 3.1)

$$\begin{aligned} u_x(x,t) &= \hat{u}_x(x,t) \\ u_y(y,t) &= \hat{u}_y(y,t) && \text{For displacements} \\ w_x(x,t) &= \hat{w}_x(x,t) && \text{on 'S}_1\text{'} \quad \dots (3.7a) \\ w_y(y,t) &= \hat{w}_y(y,t) \end{aligned}$$

$$\begin{aligned} \sigma_x \ell + \tau_{xy} m &= \hat{T}_x(x,t) \\ \tau_{xy} \ell + \sigma_y m &= \hat{T}_y(y,t) && \text{For forces on} \\ p \ell &= \hat{p}_x(x,t) && \text{'S}_2 \quad \dots (3.7b) \\ p m &= \hat{p}_y(y,t) \end{aligned}$$

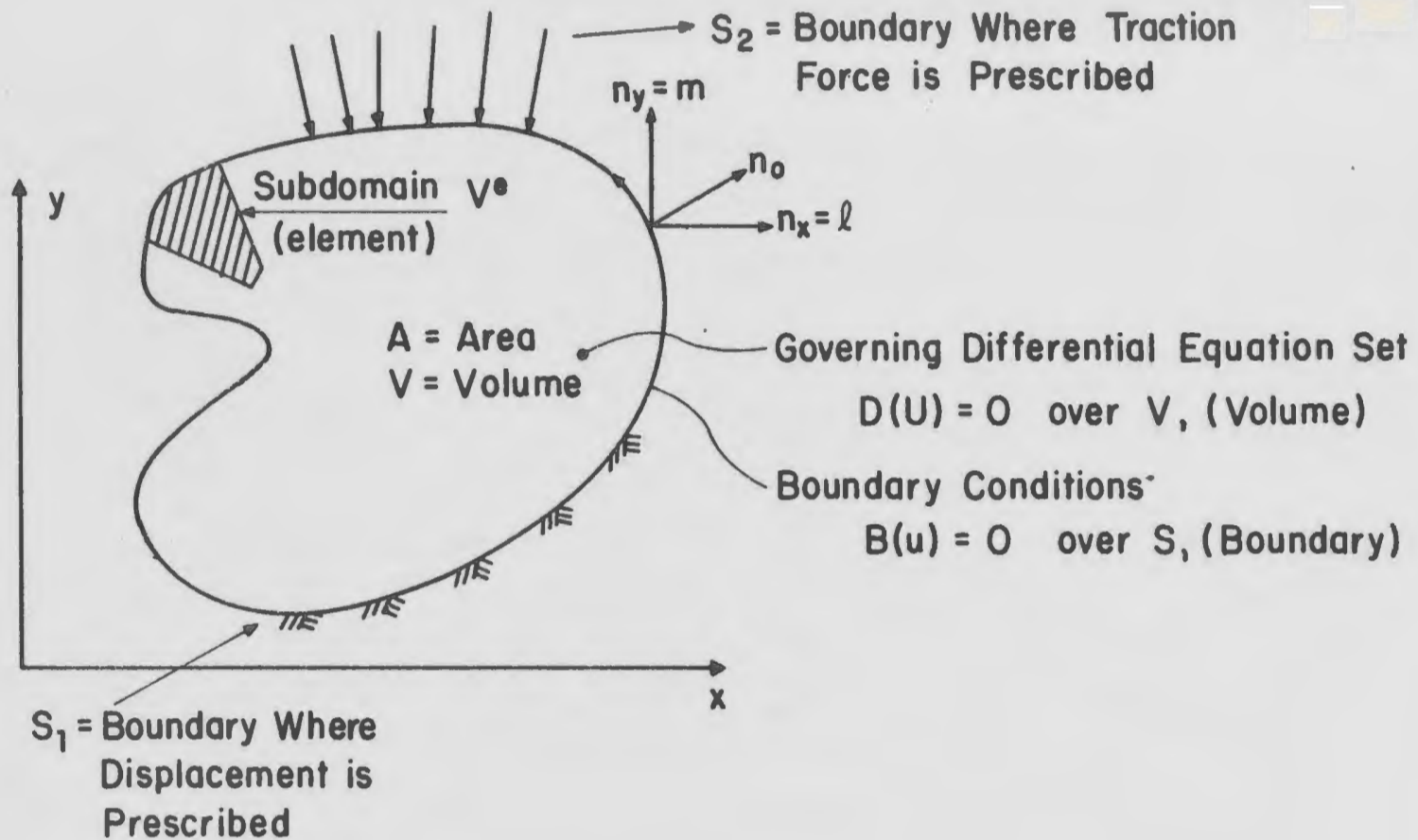


FIG. 3.1 GENERAL BOUNDARY VALUE PROBLEM

where $\hat{u}_x, \hat{u}_y =$ prescribed displacement - for bulk solid on S_1

$\hat{w}_x, \hat{w}_y =$ prescribed displacement for fluid with respect to solid on S_1

$\hat{T}_x(x,t)$ & $\hat{T}_y(y,t) =$ prescribed traction force on S_2 in x and y direction.

$\hat{p}_x(x,t)$ & $\hat{p}_y(y,t) =$ pressure on S_2 in x and y direction.

l and m are the direction cosines in x and y directions.

Applying the principle of virtual work to equations of motion following Haldar (1978) and Haldar and Reddy (1979)

$$\begin{aligned} \delta W = & \int_V \left\{ \left(\frac{\partial \sigma_x}{\partial x} + \frac{\partial \tau_{xy}}{\partial y} + \rho b_x - \rho \ddot{u}_x - \rho_f \ddot{w}_x \right) \delta u_x \right\} dV + \\ & \int_V \left\{ \left(\frac{\partial \tau_{xy}}{\partial x} + \frac{\partial \sigma_y}{\partial y} + \rho b_y - \rho \ddot{u}_y - \rho_f \ddot{w}_y \right) \delta u_y \right\} dV + \\ & \int_V \left\{ \left(\frac{\partial p}{\partial x} + \rho_f b_x - \rho_f \ddot{u}_x - \frac{1}{n} \rho_f \ddot{w}_x - \frac{1}{k} \dot{w}_x \right) \delta w_x \right\} dV + \\ & \int_V \left\{ \left(\frac{\partial p}{\partial y} + \rho_f b_y - \rho_f \ddot{u}_y - \frac{1}{n} \rho_f \ddot{w}_y - \frac{1}{k} \dot{w}_y \right) \delta w_y \right\} dV = 0 \end{aligned}$$

... (3.8)

Each term of Eqn. (3.8) is integrated using Green's theorem as follows:

$$\iiint_V \phi' \frac{\partial \psi'}{\partial x} dV = - \iiint_V \frac{\partial \phi'}{\partial x} \psi' dV + \int_S \phi' \psi' \ell dS \quad \dots (3.9)$$

Where ϕ' , and ψ' are two functions and ℓ is the direction cosine in x -direction (Fig. 3.1).

Setting $\phi' = \delta u_x$ and $\psi' = \sigma_x$ in the first integral of Eqn. (3.8)

$$\begin{aligned} \int_V \frac{\partial \sigma_x}{\partial x} \delta u_x dV &= \int_V \phi' \frac{\partial \psi'}{\partial x} dV \\ &= - \int_V \frac{\partial \phi'}{\partial x} \psi' dV + \int_S \phi' \psi' \ell dS \quad \dots (3.10) \end{aligned}$$

$$\begin{aligned} &= - \int_V \frac{\partial}{\partial x} (\delta u_x) \sigma_x dV + \int_S \sigma_x (\ell \delta u_x) dS \\ &\quad \dots (3.11a) \end{aligned}$$

Similarly the remaining terms are integrated by parts as

$$\begin{aligned} \int_V \frac{\partial \tau_{xy}}{\partial y} \delta u_x dV &= - \int_V \tau_{xy} \frac{\partial}{\partial y} (\delta u_x) dV + \int_S \tau_{xy} (m \delta u_x) dS \\ &\quad \dots (3.11b) \end{aligned}$$

Similarly, for the first term of the second integral in Eqn. (3.8)

$$\int_V \frac{\partial \tau_{xy}}{\partial x} \delta u_y dV = - \int_V \tau_{xy} \frac{\partial}{\partial x} (\delta u_y) dV + \int_S \tau_{xy} (\ell \delta u_y) dS \dots (3.11c)$$

$$\int_V \frac{\partial \sigma_y}{\partial y} \delta u_y dV = - \int_V \sigma_y \frac{\partial}{\partial y} (\delta u_y) dV + \int_S \sigma_y (m \delta u_y) dS \dots (3.11d)$$

Also, the first two terms in third and fourth integral of Eqn. (3.8) become

$$\int_V \frac{\partial p}{\partial x} \delta w_x dV = - \int_V p \frac{\partial}{\partial x} (\delta w_x) dV + \int_S p (\ell \delta w_x) dS \dots (3.11e)$$

and

$$\int_V \frac{\partial p}{\partial y} \delta w_y dV = - \int_V p \frac{\partial}{\partial y} (\delta w_y) dV + \int_S p (m \delta w_y) dS \dots (3.11f)$$

Eqn. (3.8) is rewritten in a new form using the above Eqns. (3.11a - 3.11f) as

$$\begin{aligned} \delta W = & \left\{ - \int_V \sigma_x \frac{\partial}{\partial x} (\delta u_x) dV + \int_S \sigma_x (\ell \delta u_x) dS \right\} + \left\{ - \int_V \tau_{xy} \frac{\partial}{\partial y} \right. \\ & \left. (\delta u_x) dV + \int_S \tau_{xy} (m \delta u_x) dS \right\} + \int_V \{ (\rho b_x - \rho \ddot{u}_x - \rho_f \ddot{w}_x) \delta u_x \} dV \\ & + \left\{ - \int_V \tau_{xy} \frac{\partial}{\partial x} (\delta u_y) dV + \int_S \tau_{xy} (\ell \delta u_y) dS \right\} + \left\{ - \int_V \sigma_y \right. \\ & \left. \frac{\partial}{\partial y} (\delta u_y) dV + \int_S \sigma_y (m \delta u_y) dS \right\} + \int_V \{ (\rho b_y - \rho \ddot{u}_y - \rho_f \ddot{w}_y) \} \end{aligned}$$

$$\begin{aligned}
& \delta u_y \} dV + \left\{ - \int_V p \frac{\partial}{\partial x} (\delta w_x) dV + \int_S p (\ell \delta w_x) dS \right\} \\
& + \int_V \left\{ \rho_f b_x - \rho_f \ddot{u}_x - \frac{1}{n} \rho_f \ddot{w}_x - \frac{1}{k} \dot{w}_x \right\} \delta w_x dV \\
& + \left\{ - \int_V p \frac{\partial}{\partial y} (\delta w_y) dV + \int_S p (m \delta w_y) dS \right\} \\
& + \left\{ \int_V (\rho_f b_y - \rho_f \ddot{u}_y - \frac{1}{n} \rho_f \ddot{w}_y - \frac{1}{k} \dot{w}_y) \delta w_y \right\} dV \\
& \dots (3.12)
\end{aligned}$$

Using the relations expressed in Eqn. (3.5), gives the following relationships

$$\sigma_x \frac{\partial}{\partial x} (\delta u_x) = \sigma_x \delta \left(\frac{\partial u_x}{\partial x} \right) = \sigma_x \delta \epsilon_x \quad \dots (3.13a)$$

Similarly,

$$\sigma_y \frac{\partial}{\partial y} (\delta u_y) = \sigma_y \delta \left(\frac{\partial u_y}{\partial y} \right) = \sigma_y \delta \epsilon_y \quad \dots (3.13b)$$

and

$$p \frac{\partial}{\partial x} (\delta w_x) + p \frac{\partial}{\partial y} (\delta w_y) = p \delta \left(\frac{\partial w_x}{\partial x} + \frac{\partial w_y}{\partial y} \right) = p \delta \xi \quad \dots (3.13c)$$

Finally, the following relationship is obtained after substituting Eqn. (3.13) in Eqn. (3.12)

$$\delta W = - \int_V (\sigma_x \delta \epsilon_x + \sigma_y \delta \epsilon_y + \tau_{xy} \delta \gamma_{xy} + p \delta \xi) dV + \int_{S_2} \{ (\sigma_x \ell$$

$$\begin{aligned}
& + \tau_{xy}^m) \delta u_x \, dS - \int_V \{(\rho \ddot{u}_y + \rho_f \ddot{w}_y) \delta u_y\} \, dV + \int_{S_2} \{(\tau_{xy}^l \\
& + \sigma_y^m) \delta u_y\} \, dS + \int_V (\rho b_x \delta u_x) \, dV + \int_V (\rho b_y \delta u_y) \, dV - \int_V dV \delta u_x \\
& \{(\rho \ddot{u}_x + \rho_f \ddot{w}_x) + \int_{S_2} (p_l \delta w_x + p_m \delta w_y) \, dS + \int_V (\rho_f b_x \delta w_x) \, dV \\
& + \int_V (\rho_f b_y \delta w_y) \, dV - \int_V \{(\rho_f \ddot{u}_x + \frac{\rho_f}{n} \ddot{w}_x) \delta w_x\} \, dV - \int_V \frac{1}{k} \\
& (\dot{w}_x \delta w_x) \, dV - \int_V \{\rho_f \ddot{u}_y + \frac{1}{n} \rho_f \ddot{w}_y\} \delta u_y \, dV - \int_V \frac{1}{k} (\dot{w}_y \delta w_y) \, dV \\
& \dots (3.14)
\end{aligned}$$

$$\begin{aligned}
& = - \int_V \langle \delta \epsilon_x \, \delta \epsilon_y \, \delta \gamma_{xy} \, \delta \xi \rangle \begin{Bmatrix} \sigma_x \\ \sigma_y \\ \tau_{xy} \\ p_{xy} \end{Bmatrix} \, dV + \int_{S_2} \langle \delta u_x \, \delta u_y \rangle \begin{Bmatrix} \hat{T}_x \\ \hat{T}_y \end{Bmatrix} \, dS \\
& + \int_V \langle \delta u_x \, \delta u_y \rangle \begin{Bmatrix} X_B \\ Y_B \end{Bmatrix} \, dV - \int_V \langle \delta u_x \, \delta u_y \rangle \begin{Bmatrix} \rho \ddot{u}_x \\ \rho \ddot{u}_y \end{Bmatrix} \, dV - \int_V \langle \delta u_x \, \delta u_y \rangle \\
& \begin{Bmatrix} \rho_f \ddot{w}_x \\ \rho_f \ddot{w}_y \end{Bmatrix} \, dV + \int_{S_2} \langle \delta w_x \, \delta w_y \rangle \begin{Bmatrix} \hat{p}_x \\ \hat{p}_y \end{Bmatrix} \, dS - \int_V \langle \delta w_x \, \delta w_y \rangle \begin{Bmatrix} \rho_f \ddot{u}_x \\ \rho_f \ddot{u}_y \end{Bmatrix} \, dV \\
& - \int_V \langle \delta w_x \, \delta w_y \rangle \begin{Bmatrix} \frac{\rho_f}{n} \ddot{w}_x \\ \frac{\rho_f}{n} \ddot{w}_y \end{Bmatrix} \, dV - \int_V \langle \delta w_x \, \delta w_y \rangle \begin{Bmatrix} \frac{1}{k} \dot{w}_x \\ \frac{1}{k} \dot{w}_y \end{Bmatrix} \, dV \\
& + \int_V \langle \delta w_x \, \delta w_y \rangle \begin{Bmatrix} \rho_f b_x \\ \rho_f b_y \end{Bmatrix} \, dV \\
& \dots (3.15)
\end{aligned}$$

$$\begin{aligned}
& = - \int_V \delta \epsilon^T \sigma \, dV + \int_{S_2} \delta u^T \{\hat{T}\} \, dS + \int_V \delta u^T \begin{Bmatrix} X_B \\ Y_B \end{Bmatrix} \, dV \\
& - \int_V \rho \delta u^T \begin{Bmatrix} \ddot{u}_x \\ \ddot{u}_y \end{Bmatrix} \, dV - \int_V \rho_f \delta u^T \begin{Bmatrix} \ddot{w}_x \\ \ddot{w}_y \end{Bmatrix} \, dV + \int_{S_2} \delta w^T \{\hat{p}\} \, dS
\end{aligned}$$

$$\begin{aligned}
& -\int_V \delta w^T dV \rho_f \begin{Bmatrix} \ddot{u}_x \\ \ddot{u}_y \end{Bmatrix} - \int_V \frac{\rho_f}{n} \delta w^T \begin{Bmatrix} \ddot{w}_x \\ \ddot{w}_y \end{Bmatrix} dV - \int_V \frac{1}{k} \delta w^T \begin{Bmatrix} \dot{\ddot{w}}_x \\ \dot{\ddot{w}}_y \end{Bmatrix} dV \\
& - \int_V \delta w^T \begin{Bmatrix} X_F \\ Y_F \end{Bmatrix} dV \quad \dots (3.16)
\end{aligned}$$

where $\delta \epsilon^T = \langle \delta \epsilon_x \quad \delta \epsilon_y \quad \delta \gamma_{xy} \quad \delta \xi \rangle$

$$\sigma^T = \langle \sigma_x \quad \sigma_y \quad \tau_{xy} \quad p \rangle$$

and T denotes the transpose

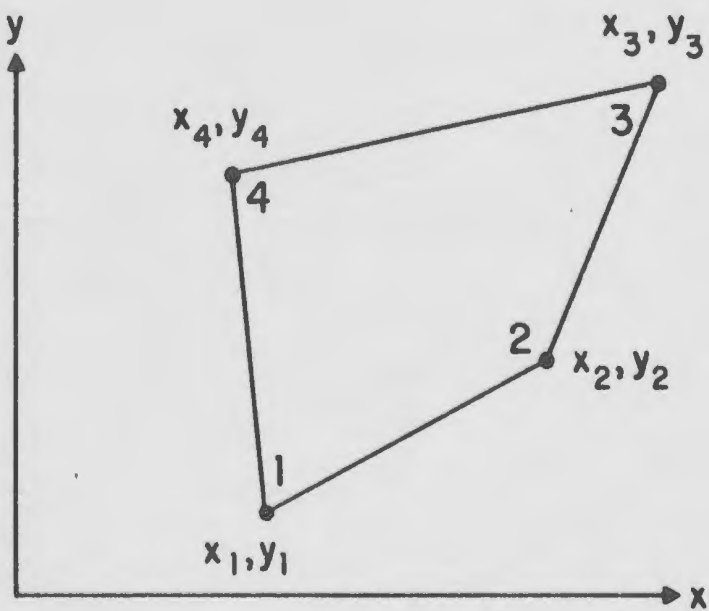
Here, S_2 , represents that part of the total surface, S , for which stresses are prescribed, and not displacements (Fig. 3.1)

3.7 Two Dimensional Finite Element Formulation

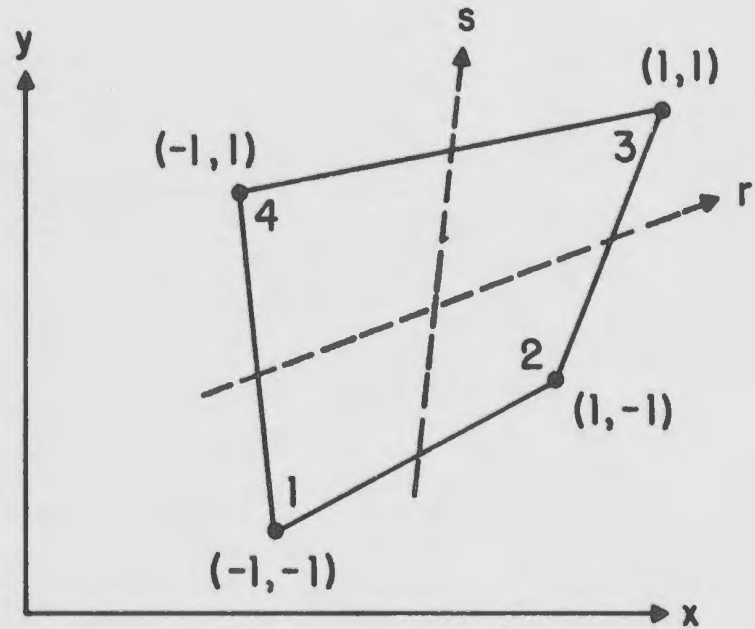
A four noded isoparametric element shown in Fig. 3.2 is used in which a linear displacement field is assumed for the solid and the fluid. The continuum is discretized using four noded elements. The soil is assumed to be homogeneous within each element.

For a four noded isoparametric element, the coordinate interpolations are assumed following Haldar and Reddy (1979) as

$$x = \sum_{i=1}^4 h_i x_i \quad \dots (3.17a)$$



a. GLOBAL SYSTEM



b. LOCAL SYSTEM

FIG. 3.2 TWO-DIMENSIONAL ISOPARAMETRIC PLANE STRAIN ELEMENT

$$y = \sum_{i=1}^4 h_i y_i \quad \dots (3.17b)$$

The displacement interpolation functions for the solid and fluid are

$$u_x = \sum_{i=1}^4 h_i U_{xi} \quad \dots (3.18a)$$

$$u_y = \sum_{i=1}^4 h_i U_{yi} \quad \dots (3.18b)$$

and

$$w_x = \sum_{i=1}^4 h_i W_{xi} \quad \dots (3.18c)$$

$$w_y = \sum_{i=1}^4 h_i W_{yi} \quad \dots (3.18d)$$

where

$$\begin{aligned} h_1 &= \frac{1}{4} (1 + r) (1 + s) \\ h_2 &= \frac{1}{4} (1 - r) (1 + s) \\ h_3 &= \frac{1}{4} (1 - r) (1 - s) \\ h_4 &= \frac{1}{4} (1 + r) (1 - s) \end{aligned} \quad \dots (3.19)$$

Here, r, s , and x, y represent the local and global coordinate systems respectively (Fig. 3.2). The solid displacements, within an element are expressed in terms of nodal displacements vectors $\underline{U}_x = (U_{x1}, U_{x2}, U_{x3}, U_{x4})$ and $\underline{U}_y = (U_{y1}, U_{y2}, U_{y3}, U_{y4})$ as

$$\begin{aligned}
 \begin{bmatrix} u_x \\ u_y \end{bmatrix} &= \begin{bmatrix} h_1 & h_2 & h_3 & h_4 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & h_1 & h_2 & h_3 & h_4 \end{bmatrix} \begin{bmatrix} U_x \\ U_y \end{bmatrix} \\
 &= \begin{bmatrix} \phi_m & 0 \\ 0 & \phi_m \end{bmatrix} \begin{bmatrix} U_x \\ U_y \end{bmatrix} \quad \dots (3.20a)
 \end{aligned}$$

$$\text{or } u = \phi_m \underline{U} ,$$

where u is solid displacement vector at any point within the element

and \underline{U} is displacement vector at nodal points

Similarly, the fluid displacement with respect to solid within an element can also be rewritten in terms of fluid nodal displacement vectors as

$$\begin{aligned}
 \begin{bmatrix} w_x \\ w_y \end{bmatrix} &= \begin{bmatrix} h_1 & h_2 & h_3 & h_4 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & h_1 & h_2 & h_3 & h_4 \end{bmatrix} \begin{bmatrix} W_x \\ W_y \end{bmatrix} \\
 &= \begin{bmatrix} \phi_m & 0 \\ 0 & \phi_m \end{bmatrix} \begin{bmatrix} W_x \\ W_y \end{bmatrix} \quad \dots (3.20b)
 \end{aligned}$$

$$\text{or, } w = \phi_m \underline{W}$$

where w is fluid displacement vector at any point within the element

and \underline{W} is displacement vector at nodal points

From Eqns. (3.20a) and (3.20b) the following

relationships are obtained:

$$\underline{u}^T = \underline{U}^T \underline{\phi}_m^T \quad \text{and} \quad \underline{w}^T = \underline{W}^T \underline{\phi}_m^T \quad \dots (3.21a)$$

$$\ddot{\underline{u}} = \underline{\phi}_m \ddot{\underline{U}} \quad \text{and} \quad \ddot{\underline{w}} = \underline{\phi}_m \ddot{\underline{W}} \quad \dots (3.21b)$$

$$\dot{\underline{u}} = \underline{\phi}_m \dot{\underline{U}} \quad \text{and} \quad \dot{\underline{w}} = \underline{\phi}_m \dot{\underline{W}} \quad \dots (3.21c)$$

where $\dot{\underline{u}}$, $\ddot{\underline{u}}$ = velocity and acceleration vectors for solid

and $\dot{\underline{w}}$, $\ddot{\underline{w}}$ = velocity and acceleration vectors for fluid with respect to solid.

The strain displacement relationships given by Eqn. (3.5) are established as follows in terms of the derivatives of the interpolation function as

$$\epsilon = \begin{bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \\ \xi \end{bmatrix} = \begin{bmatrix} \frac{\partial u_x}{\partial x} \\ \frac{\partial u_y}{\partial y} \\ \frac{\partial u_x}{\partial y} + \frac{\partial u_y}{\partial x} \\ \frac{\partial w_x}{\partial x} + \frac{\partial w_y}{\partial y} \end{bmatrix} = \begin{bmatrix} \frac{\partial}{\partial x} & 0 & 0 & 0 \\ 0 & \frac{\partial}{\partial y} & 0 & 0 \\ \frac{\partial}{\partial y} & \frac{\partial}{\partial x} & 0 & 0 \\ 0 & 0 & \frac{\partial}{\partial x} & \frac{\partial}{\partial y} \end{bmatrix} \begin{bmatrix} u_x \\ u_y \\ w_x \\ w_y \end{bmatrix} \quad \dots (3.22)$$

After substituting Eqns. (3.20a and 3.20b), Eqn. (3.22) becomes

$$\epsilon = \begin{bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \\ \xi \end{bmatrix} = \begin{bmatrix} \frac{\partial}{\partial x} & 0 & 0 & 0 \\ 0 & \frac{\partial}{\partial y} & 0 & 0 \\ \frac{\partial}{\partial y} & \frac{\partial}{\partial x} & 0 & 0 \\ 0 & 0 & \frac{\partial}{\partial x} & \frac{\partial}{\partial y} \end{bmatrix} \begin{bmatrix} \phi_m & 0 & 0 & 0 \\ 0 & \phi_m & 0 & 0 \\ 0 & 0 & \phi_m & 0 \\ 0 & 0 & 0 & \phi_m \end{bmatrix} \begin{bmatrix} U_x \\ U_y \\ W_x \\ W_y \end{bmatrix} \quad \dots (3.23)$$

$$= \begin{bmatrix} \frac{\partial}{\partial x} (\phi_m) & 0 & 0 & 0 \\ 0 & \frac{\partial}{\partial y} (\phi_m) & 0 & 0 \\ \frac{\partial}{\partial y} (\phi_m) & \frac{\partial}{\partial x} (\phi_m) & 0 & 0 \\ 0 & 0 & \frac{\partial}{\partial x} (\phi_m) & \frac{\partial}{\partial y} (\phi_m) \end{bmatrix} \begin{bmatrix} U_x \\ U_y \\ W_x \\ W_y \end{bmatrix} \quad \dots (3.24)$$

$$= \begin{bmatrix} \psi_{11} & \psi_{12} \\ \psi_{21} & \psi_{22} \end{bmatrix} \begin{bmatrix} U \\ W \end{bmatrix} = \begin{bmatrix} \psi \\ \psi \end{bmatrix} \begin{bmatrix} U \\ W \end{bmatrix} \quad \dots (3.25)$$

where

$$\psi_{11} = \begin{bmatrix} \frac{\partial}{\partial x} (\underline{\phi}_m) & 0 \\ 0 & \frac{\partial}{\partial y} (\underline{\phi}_m) \\ \frac{\partial}{\partial y} (\underline{\phi}_m) & \frac{\partial}{\partial x} (\underline{\phi}_m) \end{bmatrix} \quad \dots (3.25a)$$

and

$$\underline{U} = \begin{bmatrix} U_x \\ U_y \end{bmatrix}, \quad \underline{W} = \begin{bmatrix} W_x \\ W_y \end{bmatrix} \quad \dots (3.25b)$$

$$\psi_{12} = \begin{bmatrix} 0 & 0 \\ 0 & 0 \\ 0 & 0 \end{bmatrix} \quad \dots (3.25c)$$

$$\psi_{21} = \begin{bmatrix} 0 & 0 \end{bmatrix} \quad \dots (3.25d)$$

$$\psi_{22} = \begin{bmatrix} \frac{\partial}{\partial x} (\underline{\phi}_m) & \frac{\partial}{\partial y} (\underline{\phi}_m) \end{bmatrix} \quad \dots (3.25e)$$

The stress-strain relation can be rewritten from Eqns. (3.6e) in the following form

$$\begin{bmatrix} \sigma_x \\ \sigma_y \\ \tau_{xy} \\ p \end{bmatrix} = \begin{bmatrix} \lambda_c + 2\mu & \lambda_c & 0 & \alpha M_o \\ \lambda_c & \lambda_c + 2\mu & 0 & \alpha M_o \\ 0 & 0 & \mu & \alpha M_o \\ \alpha M_o & \alpha M_o & 0 & M_o \end{bmatrix} \begin{bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \\ \xi \end{bmatrix}$$

$$\text{or, } \sigma = \begin{bmatrix} \sigma_s \\ p \end{bmatrix} = \begin{bmatrix} D_{ss} & D_{sf} \\ D_{fs} & D_{ff} \end{bmatrix} \begin{bmatrix} \epsilon_s \\ \xi \end{bmatrix} \quad \dots (3.26)$$

where D_{ss} = solid stress-strain matrix (undrained)

$D_{sf} = D_{fs}$ = solid-fluid coupling stress-strain matrix

D_{ff} = fluid stress-strain matrix

and $\sigma_s^T = \langle \sigma_x \quad \sigma_y \quad \tau_{xy} \rangle$

$\epsilon_s^T = \langle \epsilon_x \quad \epsilon_y \quad \gamma_{xy} \rangle$

Substitution of Eqn. (3.25) in stress-strain relation of Eqn. (3.26) furnishes

$$\sigma = \begin{bmatrix} D_{ss} & D_{sf} \\ D_{fs} & D_{ff} \end{bmatrix} \begin{bmatrix} \psi_{11} & \psi_{12} \\ \psi_{21} & \psi_{22} \end{bmatrix} \begin{bmatrix} \underline{U} \\ \underline{W} \end{bmatrix} \quad \dots (3.27)$$

or

$$\sigma = \begin{bmatrix} \underline{D} \\ \underline{\psi} \end{bmatrix} \begin{bmatrix} \underline{U} \\ \underline{W} \end{bmatrix} \\ = \begin{bmatrix} D_{ss}\psi_{11} & (D_{ss}\psi_{12} + D_{sf}\psi_{22}) \\ D_{fs}\psi_{11} & D_{ff}\psi_{22} \end{bmatrix} \begin{bmatrix} \underline{U} \\ \underline{W} \end{bmatrix} \quad \dots (3.28)$$

where $D_{ss}\psi_{12} = 0$

From Eqn. (3.25) the strain displacement relation, is obtained in partitioned form as

$$\langle \epsilon_x \quad \epsilon_y \quad \gamma_{xy} \rangle = \underline{U}^T \psi_{11}^T$$

$$\xi = \underline{W}^T \psi_{22}^T \quad \dots (3.29)$$

or,

$$\langle \delta \epsilon_x \quad \delta \epsilon_y \quad \delta \gamma_{xy} \rangle = \delta \underline{U}^T \psi_{11}^T \quad \dots (3.30)$$

and

$$\delta \xi = \delta \underline{W}^T \psi_{22}^T \quad \dots (3.31)$$

Rewriting Eqn. (3.27) in the following form

$$\sigma = \begin{bmatrix} D_{ss} & D_{sf} \\ D_{fs} & D_{ff} \end{bmatrix} \begin{bmatrix} \psi_{11} \underline{U} \\ \psi_{22} \underline{W} \end{bmatrix} = \begin{bmatrix} D_{ss} \psi_{11} \underline{U} + D_{sf} \psi_{22} \underline{W} \\ D_{fs} \psi_{11} \underline{U} + D_{ff} \psi_{22} \underline{W} \end{bmatrix} \quad \dots (3.32)$$

and substitution of the above relations in Eqn.

(3.16) gives

$$\begin{aligned} \delta W = & - \int_V \delta \underline{U}^T \psi_{11}^T (D_{ss} \psi_{11} \underline{U} + D_{sf} \psi_{22} \underline{W}) dV - \int_V \delta \underline{W}^T \psi_{22}^T \\ & (D_{fs} \psi_{11} \underline{U} + D_{ff} \psi_{22} \underline{W}) dV + \int_{S_2} \delta \underline{U}^T \underline{\phi}_m^T \{\hat{T}\} dS \\ & + \int_V \delta \underline{U}^T \underline{\phi}_m^T \begin{Bmatrix} X_B \\ Y_B \end{Bmatrix} dV - \int_V \rho \delta \underline{U}^T \underline{\phi}_m^T \underline{\phi}_m \ddot{\underline{U}} dV \end{aligned}$$

$$\begin{aligned}
& - \int_V \rho_f \delta \underline{U}^T \underline{\phi}_m^T \underline{\phi}_m \ddot{\underline{W}} \, dV + \int_{S_2} \delta \underline{W}^T \underline{\phi}_m^T \{\hat{p}\} \, dS \\
& - \int_V \delta \underline{W}^T \rho_f \underline{\phi}_m^T \underline{\phi}_m \ddot{\underline{U}} \, dV - \int_V \delta \underline{W}^T \underline{\phi}_m^T \frac{\rho_f}{n} \underline{\phi}_m \ddot{\underline{W}} \, dV \\
& - \int_V \delta \underline{W}^T \underline{\phi}_m^T \frac{1}{k} \underline{\phi}_m \dot{\underline{W}} \, dV + \int_V \delta \underline{W}^T \underline{\phi}_m^T \begin{Bmatrix} X_F \\ Y_F \end{Bmatrix} \, dV \\
& \dots (3.33)
\end{aligned}$$

As $\delta W = 0$ and $\delta \underline{U}^T$ and $\delta \underline{W}^T$ are arbitrary variations in displacements for solids and fluids with respect to solids, two sets of equations, which are independently zero, are obtained as:

$$\begin{aligned}
& \left(\int_V \underline{\phi}_m^T \rho_f \underline{\phi}_m \ddot{\underline{U}} + \int_V \underline{\phi}_m^T \rho_f \underline{\phi}_m \ddot{\underline{W}} + \int_V \psi_{11}^T D_{ss} \psi_{11} \underline{U} + \int_V \psi_{11}^T D_{sf} \right. \\
& \left. \psi_{22} \underline{W} \right) \, dV = \int_V \underline{\phi}_m^T \begin{Bmatrix} X_B \\ Y_B \end{Bmatrix} \, dV + \int_{S_2} \underline{\phi}_m^T \{\hat{T}\} \, dS \\
& \dots (3.34a)
\end{aligned}$$

and

$$\begin{aligned}
& \left(\int_V \underline{\phi}_m^T \rho_f \underline{\phi}_m \ddot{\underline{U}} + \int_V \underline{\phi}_m^T \frac{\rho_f}{n} \underline{\phi}_m \ddot{\underline{W}} + \int_V \underline{\phi}_m^T \frac{1}{k} \underline{\phi}_m \dot{\underline{W}} + \int_V \right. \\
& \left. \psi_{22}^T D_{sf} \psi_{11} \underline{U} + \int_V \psi_m^T D_{ff} \psi_{22} \underline{W} \right) \, dV = \int_V \underline{\phi}_m^T \begin{Bmatrix} X_F \\ Y_F \end{Bmatrix} \, dV \\
& + \int_{S_2} \underline{\phi}_m^T \{\hat{p}\} \, dS \quad \dots (3.34b)
\end{aligned}$$

or

$$\begin{bmatrix} M_{ss} & M_{sf} \\ M_{fs} & M_{ff} \end{bmatrix} \begin{bmatrix} \ddot{\underline{U}} \\ \ddot{\underline{W}} \end{bmatrix} + \begin{bmatrix} \underline{0} & \underline{0} \\ \underline{0} & C_{ff} \end{bmatrix} \begin{bmatrix} \dot{\underline{U}} \\ \dot{\underline{W}} \end{bmatrix} + \begin{bmatrix} K_{ss} & K_{sf} \\ K_{fs} & K_{ff} \end{bmatrix} \begin{bmatrix} \underline{U} \\ \underline{W} \end{bmatrix} = \begin{bmatrix} \underline{P}_B \\ \underline{P}_f \end{bmatrix} \quad \dots (3.35)$$

The structural damping which accounts for the energy dissipation of solid skeleton, may be represented in terms of Rayleigh damping coefficients (B_1 and B_2) as a combination of mass and stiffness matrices and expressed in the following form as

$$C_{ss} = B_1 (M_{ss} - n^2 M_{ff}) + B_2 (K_{ss} - \alpha^2 K_{ff}) \quad \dots (3.36)$$

Substitution of Eqn. (3.36) in (3.35) gives

$$\begin{bmatrix} M_{ss} & M_{sf} \\ M_{fs} & M_{ff} \end{bmatrix} \begin{bmatrix} \ddot{\underline{U}} \\ \ddot{\underline{W}} \end{bmatrix} + \begin{bmatrix} C_{ss} & \underline{0} \\ \underline{0} & C_{ff} \end{bmatrix} \begin{bmatrix} \dot{\underline{U}} \\ \dot{\underline{W}} \end{bmatrix} + \begin{bmatrix} K_{ss} & K_{sf} \\ K_{fs} & K_{ff} \end{bmatrix} \begin{bmatrix} \underline{U} \\ \underline{W} \end{bmatrix} = \begin{bmatrix} \underline{P}_B \\ \underline{P}_f \end{bmatrix} \quad \dots (3.37)$$

where

M_{ss} = solid consistent mass matrix

$$= \int_V \underline{\phi}_m^T \rho \underline{\phi}_m dV, \quad \dots (3.38a)$$

$$\begin{aligned}
 M_{sf} &= \text{coupled solid fluid matrix} \\
 &= \int_V \underline{\phi}_m^T \rho_f \underline{\phi}_m dV , \quad \dots (3.38b)
 \end{aligned}$$

$$\begin{aligned}
 M_{ff} &= \text{fluid consistent mass matrix} \\
 &= \int_V \underline{\phi}_m^T \frac{\rho_f}{n} \underline{\phi}_m dV , \quad \dots (3.38c)
 \end{aligned}$$

$$\begin{aligned}
 C_{ff} &= \text{fluid damping matrix} \\
 &= \int_V \underline{\phi}_m^T \frac{1}{k} \underline{\phi}_m dV , \quad \dots (3.38d)
 \end{aligned}$$

$$\begin{aligned}
 K_{ss} &= \text{solid stiffness matrix (undrained)} \\
 &= \int_V \psi_{11}^T D_{ss} \psi_{11} dV , \quad \dots (3.38e)
 \end{aligned}$$

$$\begin{aligned}
 K_{sf} &= K_{fs} = \text{solid fluid coupled stiffness} \\
 &\quad \text{matrix} \\
 &= \int_V \psi_{11}^T D_{sf} \psi_{22} = \int_V \psi_{22}^T D_{fs} \psi_{11} dV \quad \dots (3.38f)
 \end{aligned}$$

$$\begin{aligned}
 K_{ff} &= \text{fluid stiffness matrix} \\
 &= \int_V \psi_{22}^T D_{ff} \psi_{22} dV , \quad \dots (3.38g)
 \end{aligned}$$

Using Eqn. (3.6e), the term, D_{ss} , in Eqn. (3.38e) can further be rewritten as

$$K_{ss} = \int_V \psi_{11}^T [D_{sd} + D_{ff}] \psi_{11} dV \quad \dots (3.38h)$$

where D_{sd} = drained elasticity matrix for the soil skeleton

$$= \begin{bmatrix} \lambda+2\mu & \lambda & 0 \\ \lambda & \lambda+2\mu & 0 \\ 0 & 0 & \mu \end{bmatrix}$$

and D_{ff} = elasticity matrix for the pore fluid

$$= \begin{bmatrix} \alpha^2 M_o & \alpha^2 M_o & 0 \\ \alpha^2 M_o & \alpha^2 M_o & 0 \\ 0 & 0 & 0 \end{bmatrix} \quad \dots (3.38i)$$

Substituting Eqn. (3.25a) in Eqn. (3.38h) one can obtain

$$K_{ss} = \int_V \psi_{11}^T D_{sd} \psi_{11} dV + \int_V \alpha^2 M_o \psi_{22}^T \psi_{22} dV \quad \dots (3.38j)$$

where $\psi_{22} = [h_{1,x}, h_{2,x}, \dots, h_{3,y}, h_{4,y}]$

and $h_{i,x}$ = differentiation of the i -th interpolaton function with respect to x -coordinate

The load vectors in Eqn. (3.37) are expressed as

$$\begin{aligned}
 p_B &= \text{bulk load vector including gravity} \\
 &= \int_V \phi_m^T \begin{Bmatrix} X_B \\ Y_B \end{Bmatrix} dV + \int_{S_2} \phi_m^T \{\hat{T}\} dS \\
 &\dots (3.381)
 \end{aligned}$$

and

$$\begin{aligned}
 p_f &= \text{fluid load vector including gravity} \\
 &= \int_V \phi_m^T \begin{Bmatrix} X_F \\ Y_F \end{Bmatrix} dV + \int_{S_2} \phi_m^T \{\hat{p}\} dS \\
 &\dots (3.38m)
 \end{aligned}$$

3.8 Condition for Incompressibility

The redistribution of the stresses takes place just after the loading is applied to the soil medium. The stresses here are termed as total and effective and are accompanied by strains and displacements. The pore water pressure will also develop and the magnitude and distribution of the pore water pressure depends not only on the load but also on the soil type and its past stress history. If the loading is applied in an instantaneous manner the soil behaves under undrained condition, since the pore water has little time to dissipate. The condition of undrained state for fully saturated soil is

accompanied by the condition of 'no volume' change and as such may be expressed in terms of strain components as

$$\epsilon_x + \epsilon_y + \epsilon_z = 0 \quad \dots (3.39)$$

The strain components are

$$\epsilon_x = \frac{\sigma_x}{E} - \nu \frac{\sigma_y}{E} - \nu \frac{\sigma_z}{E} \quad \dots (3.40a)$$

$$\epsilon_y = \frac{\sigma_y}{E} - \nu \frac{\sigma_x}{E} - \nu \frac{\sigma_z}{E} \quad \dots (3.40b)$$

$$\epsilon_z = \frac{\sigma_z}{E} - \nu \frac{\sigma_x}{E} - \nu \frac{\sigma_y}{E} \quad \dots (3.40c)$$

and $\gamma_{xy} = \frac{\tau_{xy}}{\mu}$ where $\nu =$ Poisson's ratio

Substitution of Eqns. (3.40a-c) into Eqn. (3.39) yields

$$(1-2\nu) \frac{\sigma_x}{E} + (1-2\nu) \frac{\sigma_y}{E} + (1-2\nu) \frac{\sigma_z}{E} = 0 \quad \dots (3.41)$$

Since σ_x , σ_y and σ_z are not equal to zero and hence $(1-2\nu) = 0$ which results in $\nu = 0.50$

The conventional displacement based finite element formulation cannot be applied to the solution of undrained deformation of soil because of the difficulty in expressing the stresses solely in terms of the

strains. Even when the finite element procedure is applied to nearly incompressible solid ($\nu \approx 0.5$) the solution will lead to very large errors.

However, the elasticity matrix, $[\underline{D}]$, in Eqn. (3.28) can be separated in the following form for undrained condition since $\xi = 0$ from Eqn. (3.5d) indicating no flow of fluid with respect to solid.

$$[\underline{D}] = (K_C - \frac{2}{3}\mu) \begin{bmatrix} 1 & 1 & 0 \\ 1 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix} + \mu \begin{bmatrix} 2 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{bmatrix} \dots (3.42)$$

where the shear modulus μ and the bulk modulus K_C represent the elasticity constants in terms of total stress, σ_{ij} , and are defined as

$$\mu = \frac{E_u}{2(1 + \nu_u)} = \frac{E_d}{2(1 + \nu_d)} \dots (3.43a)$$

and
$$K_C = \frac{E_u}{3(1 - 2\nu_u)} \dots (3.43b)$$

in which E_u and E_d = undrained and drained Young's moduli respectively.

ν_u and ν_d = undrained and drained Poisson's ratios respectively.

Eqn. (3.42) can be rewritten as

$$[D] = \mu [D]^S + 2\theta [D]^V \quad \dots (3.44)$$

where $[D]^S$ = constitutive matrix related to deviatoric component of stress-strain part

and $[D]^V$ = constitutive matrix related to volumetric component of stress-strain part

As $\nu_\mu \rightarrow 0.5$, μ remains finite when the term $2\theta \equiv (K_C - \frac{2}{3}\mu) \rightarrow \infty$

The strain energy, S^e , can then be expressed as

$$S^e = \frac{1}{2} \int_V \epsilon^T [D]^S \mu \epsilon \, dV + \frac{1}{2} \int_V \epsilon^T [D]^V 2\theta \epsilon \, dV \quad \dots (3.45)$$

The volumetric strain, ϵ_v , is expressed as

$$\epsilon_v = \epsilon_x + \epsilon_y = \frac{\partial u_x}{\partial x} + \frac{\partial u_y}{\partial y} \quad \dots (3.46)$$

$$= [1 \quad 1 \quad 0] \begin{bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \end{bmatrix} \quad \dots (3.47a)$$

and

$$S^e = S_s + S_v = \frac{1}{2} \int_V \epsilon^T [D]^S \mu \epsilon \, dV + \int_V \theta \epsilon_v^2 \, dV \quad \dots (3.47b)$$

Therefore, the term S_s represents the distortional energy whereas, the term S_v , represents the volumetric strain energy. The incompressibility condition is satisfied when $\theta \rightarrow \infty$.

In the undrained condition, the fluid has no movement with respect to the solid ie. $\underline{W} = 0$ and Eqn. (3.37) reduces to

$$[M_{ss}] \{\ddot{\underline{U}}\} + [C_{ss}] \{\dot{\underline{U}}\} + [K_{ss}] \{\underline{U}\} = \{P_B\} \quad \dots (3.48)$$

for the dynamical condition. For the quasistatic condition for undrained loading $\ddot{\underline{U}} = \dot{\underline{U}} = 0$ and Eqn. (3.48) after assembling reduces to

$$[K^*] \{\underline{U}\} = \{P_B\} \quad \dots (3.49)$$

and can be expressed in an explicit form as [Eqn. (3.38j)]

$$\int_V \psi_{11}^T D_{sd} \psi_{11} dV + \int_V \alpha^2 M_o \psi_{22}^T \psi_{22} dV \{\underline{U}\} = \{P_B\} \quad \dots (3.50)$$

where P_B = assembled load vector. We write above as

$$(K_1 + \alpha^2 M_o K_2) \{\underline{U}\} = \{P_B\} \quad \dots (3.51)$$

where K_1 and $\alpha^2 M_o K_2$ = assembled stiffness matrix for solid phase under drained

condition and that due to the effect of the degree of incompressibility of the pore fluid respectively.

The second term in Eqn. (3.51) represents the stiffness related to the volumetric strain, ϵ_v . For $\epsilon_v = 0$, $\alpha^2 M_0$ should tend to infinity under undrained condition. Eqn. (3.51) is solved with finite values for the matrices K_1 and K_2 .

It is clear that as $\alpha^2 M_0 K_2$ tends to be very large when compared to K_1 , Eqn. (3.51) becomes

$$(\alpha^2 M_0 K_2) \{ \underline{U} \} = \{ P_B \} \quad \dots (3.52)$$

and

$$[K_2] \{ \underline{U} \} = \frac{1}{\alpha^2 M_0} \{ P_B \} \rightarrow 0$$

Thus the incompressibility constraint takes a dominant role and will lead to a trivial solution if the matrix K_2 is non-singular;

Eqn. (3.52) shows the difficulty of obtaining results for the incompressible case with the displacement based finite element formulation, where the value of

the Poisson's ratio tends to one-half (0.50) but is not equal to one-half. In order to circumvent this problem, the matrix $[K_2]$ is made singular and therefore

$$\{\underline{u}\} \neq 0$$

whereas $[K_2]\{\underline{u}\} = 0$

The singularity in the $[K_2]$ matrix is introduced by use of lower order integration (Zienkiewicz, 1977) and therefore the reduced integration technique is particularly recommended for nearly incompressible problems.

The assembled undrained stiffness matrix, $[K^*]$, in global form should be non-singular, although the matrix, $[K_2]$ is singular. For linear quadrilateral element, the 2x2 integration is carried out for the matrix, $[K_1]$ and one point integration at centroid is performed for the matrix $[K_2]$. Therefore, in forming the matrix $[K_1]$ three independent relations ($\epsilon_x, \epsilon_y, \gamma_{xy}$) are introduced at each Gauss point but for the matrix $[K_2]$ only one such relation (volumetric strain, ϵ_v) is required. The condition of non-singularity of the assembled matrix, from Eqn. (3.49) is guaranteed if the number of such relations introduced at all the integrating points is greater than that of the available degrees of freedom.

3.9 Checking of Computer Programme

Two types of problems are studied in this section based on the formulation presented before. These are: a) deformation analysis of a strip footing under static vertical loading and b) dynamic response analysis of a cantilever beam under periodic loading. In both analyses, the results obtained are compared with the published values to ensure the accuracy and reliability of the computer programme.

3.9.1 Deformation Analysis of a Strip Footing

The application of the finite element formulation as described before is illustrated by the case of a strip footing underlaid by a shallow stratum of clay under undrained condition. The ratio of the half-width of the footing to the depth of the stratum is 0.50. The loading is assumed to act vertically downwards with a constant intensity and the soil medium is assumed to be homogeneous and isotropic. The finite element mesh of the strip footing is shown in Fig. (3.3). It consists of 56 nodes and 42 four noded rectangular isoparametric elements. Vertical side boundary is placed at a distance four times the width of the footing from

MESH GENERATION

NODAL POINT NUMBERING

ELEMENT NUMBERING

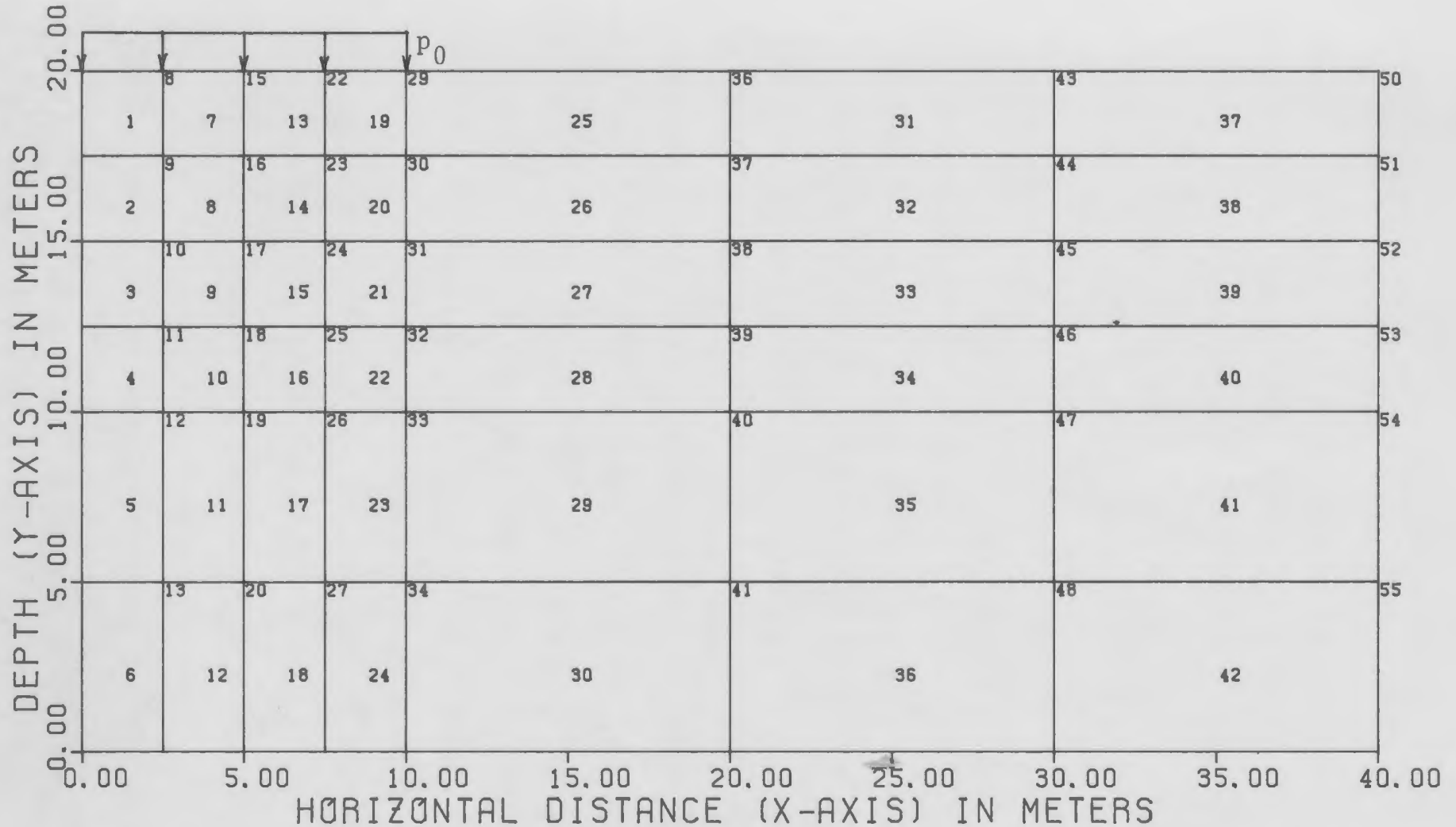


FIG. 3.3 FINITE ELEMENT MESH FOR STRIP FOUNDATION

the footing centerline and assumed to be smooth. The bottom boundary is assumed to be rough and rigid. The footing is assumed to deform under plane strain condition. Reduced integration technique is used to ensure the nearly incompressibility of the saturated soil medium.

In the initial undrained analysis the soil is modelled as a weightless two-phase material with $s_u = 100$ Kpa and $\phi_u = 0$, where s_u = undrained shear strength and ϕ_u = angle of internal friction. The drained elastic Lamé's parameters $\lambda = 150 s_u$, $\mu = 100 s_u$ and drained Poisson's ratio, $\nu_d = 0.30$ are assumed for the soil skeleton whereas the bulk modulus for pore water is assigned as $K_w = 2000$ MPa. This gives an equivalent undrained Poisson's ratio, $\nu_u \cong 0.499$ approximately ensuring the nearly incompressibility of the soil medium. For the drained analysis, K_w is only set equal to zero while the other parameters remain unaltered. The drained and undrained deformations are computed based on Eqn. (3.49) and compared with the solution presented by Booker et al (1976). Fig. 3.4 depicts the comparison of the foundation deformation at surface along horizontal direction.

Figs. 3.5 to 3.7 present the comparisons of the distribution of the stress components such as p , (pore pressure), σ_x and σ_y along the depth. The results obtained from the present analysis are in excellent agreement with those obtained by Booker et al (1976). It is interesting to note that Booker et al (1976) obtained the solution of an undrained problem based on a different approach such as defining the problem in terms of a single parameter and then expanding the solution in terms of that parameter. Figs. 3.8 and 3.9 present the stress contour plots for vertical normal stress component, σ_y , and pore pressure, p , under undrained condition whereas Fig. 3.10 depicts only the normal stress component, σ_y , for drained condition.

3.9.2 Dynamic Response Analysis of a Cantilever Beam

For the dynamic response analysis, a cantilever beam is chosen as an example problem, and is shown in Fig. 3.11. The beam is discretized with four noded isoparametric rectangular elements. The mesh consists of 3 elements and 8 nodes. Nodes 1 and 2 are fixed and the beam is subjected to equal periodic forces of $1.0 \cos(50t)$ at the free nodes 7 and 8.

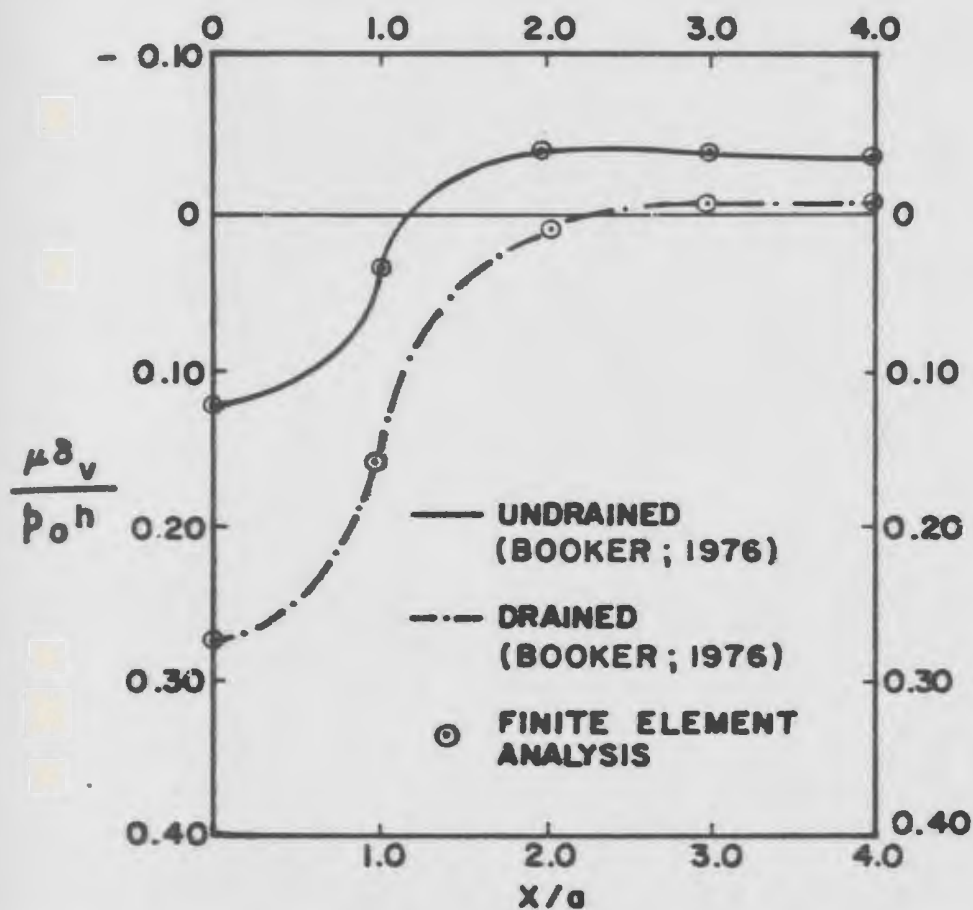


FIG. 3.4 SETTLEMENT OF THE FOUNDATION AT THE SURFACE

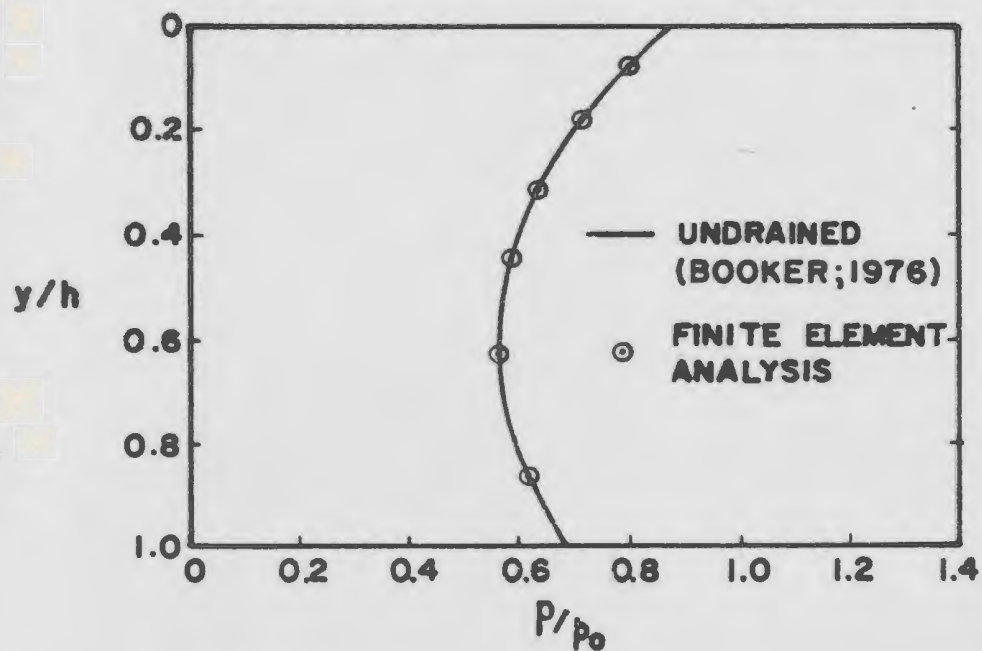


FIG. 3.5 DISTRIBUTION OF PORE PRESSURE ALONG THE CENTER LINE

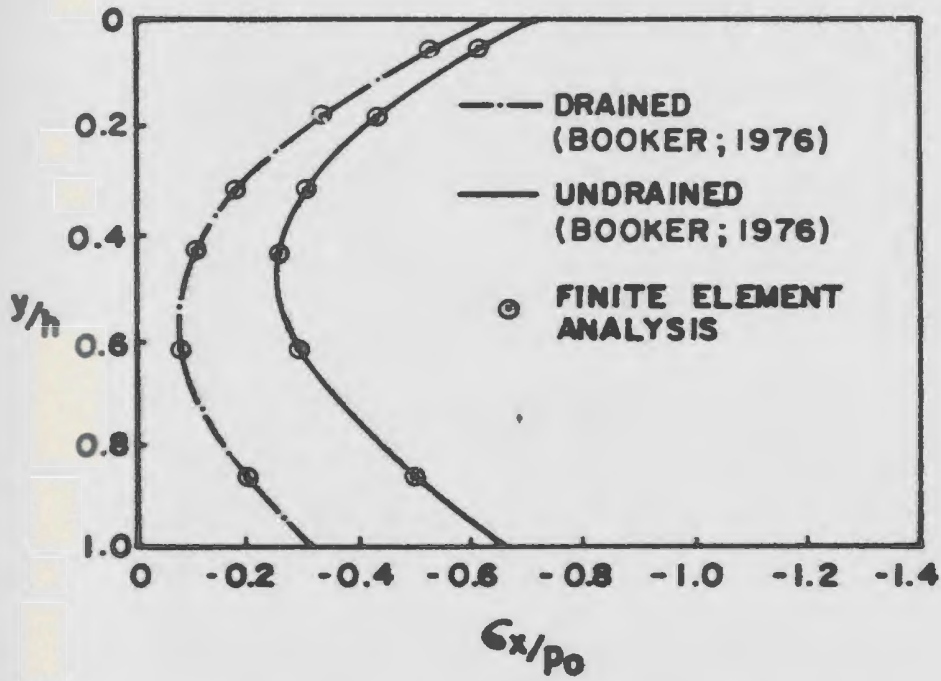


FIG. 3.6 DISTRIBUTION OF HORIZONTAL NORMAL STRESS

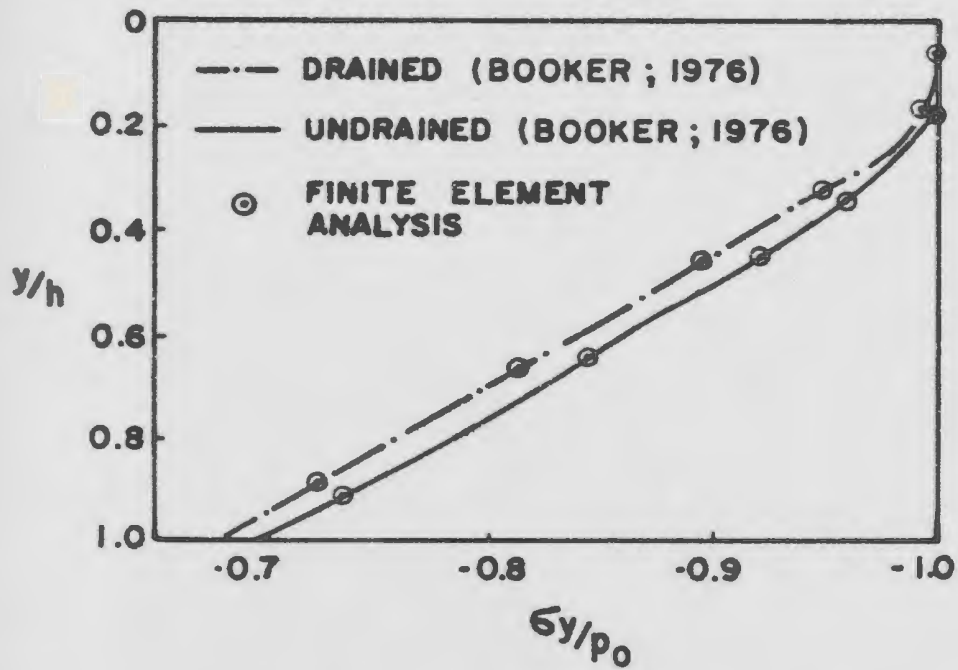


FIG. 3.7 DISTRIBUTION OF VERTICAL NORMAL STRESS

(SIGMA-Y) / (APPL. VERTICAL PRESSURE)

CONTOUR INTERVAL = 0.050

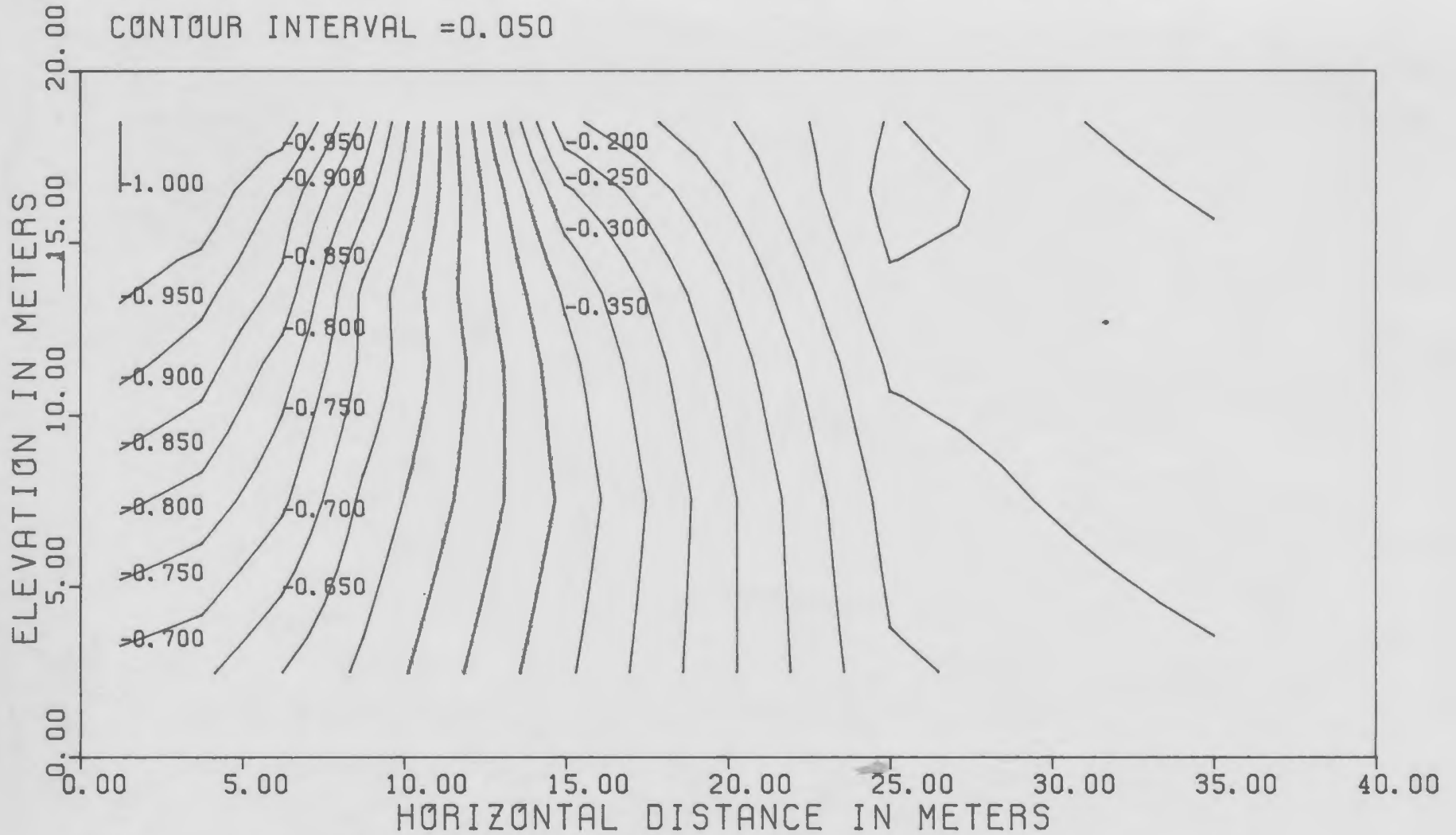


FIG. 3.8 STRESS CONTOURS FOR SIGMA-Y
(UNDRAINED CONDITION)

(PORE PRESSURE) / (APPL. VERTICAL PRESSURE)

CONTOUR INTERVAL = 0.050

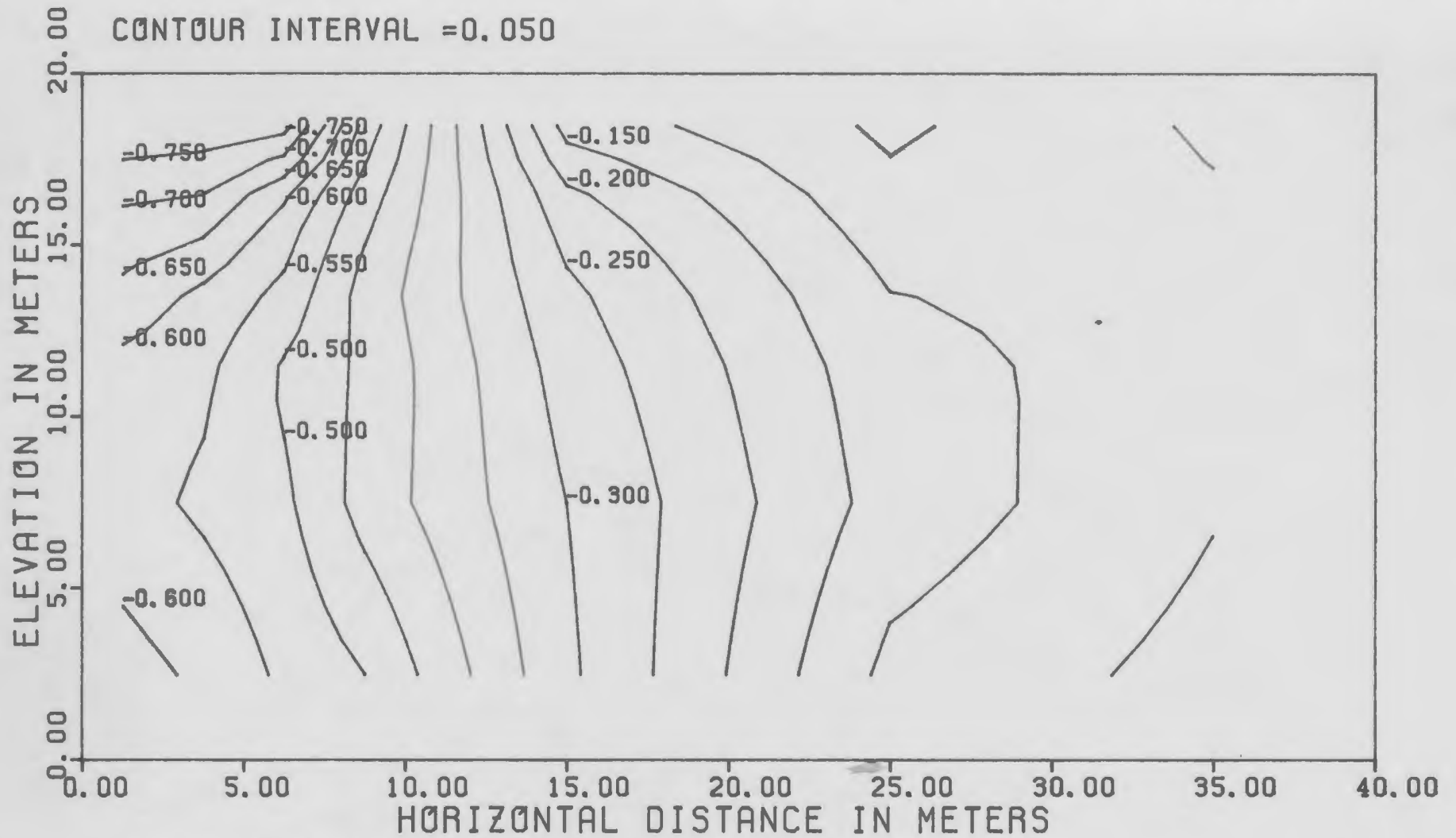


FIG. 3.9 PORE PRESSURE CONTOURS
(UNDRAINED CONDITION)

(SIGMA-Y) / (APPL. VERTICAL PRESSURE)

CONTOUR INTERVAL = 0.050

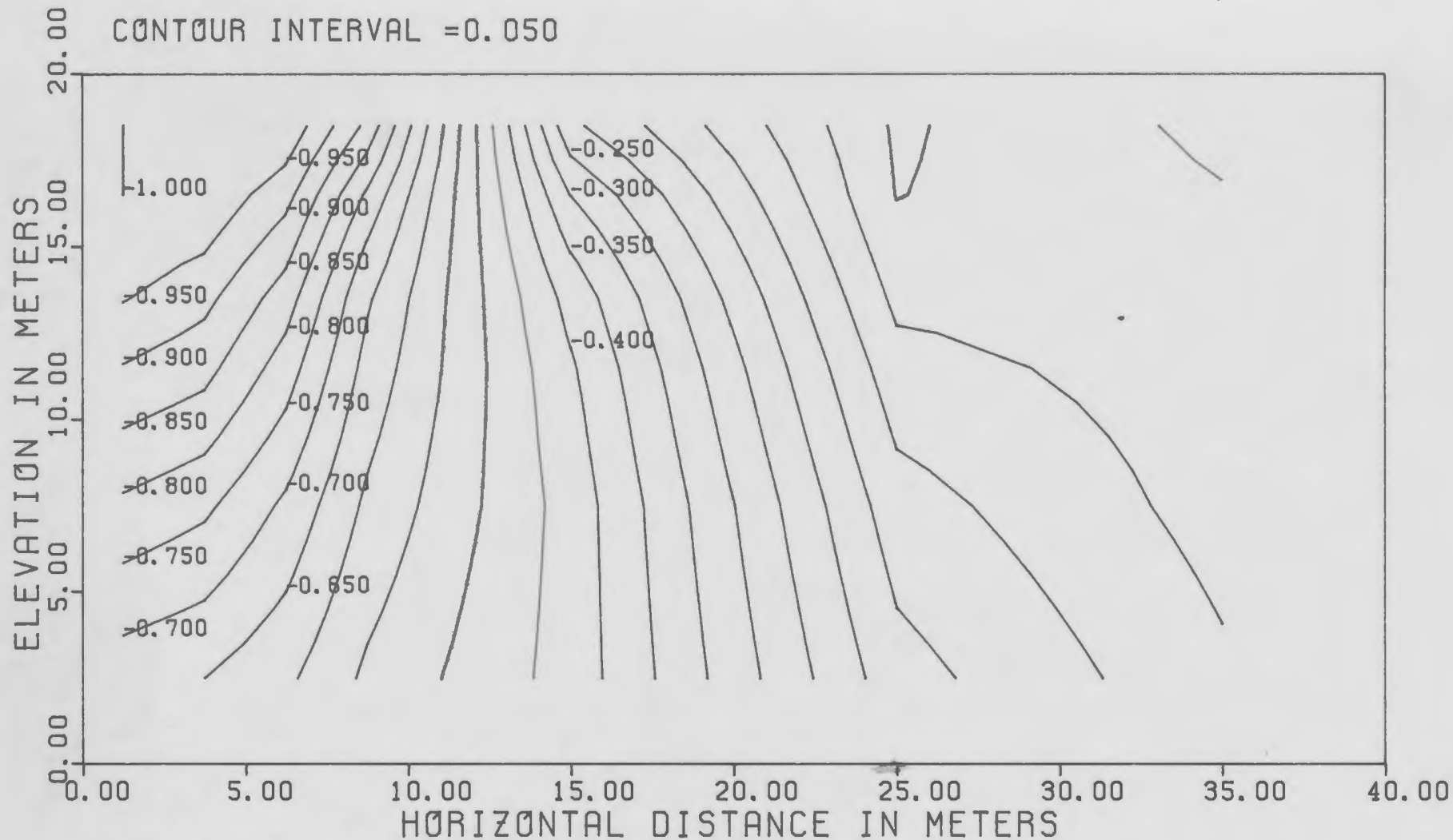


FIG. 3.10 STRESS CONTOURS FOR SIGMA-Y
(DRAINED CONDITION)

$$a = 4.0$$

$$b = 1.0$$

$$E = 1.0E + 06$$

$$\nu = 0.30$$

$$\Delta t = 0.005 \text{ SECONDS}$$

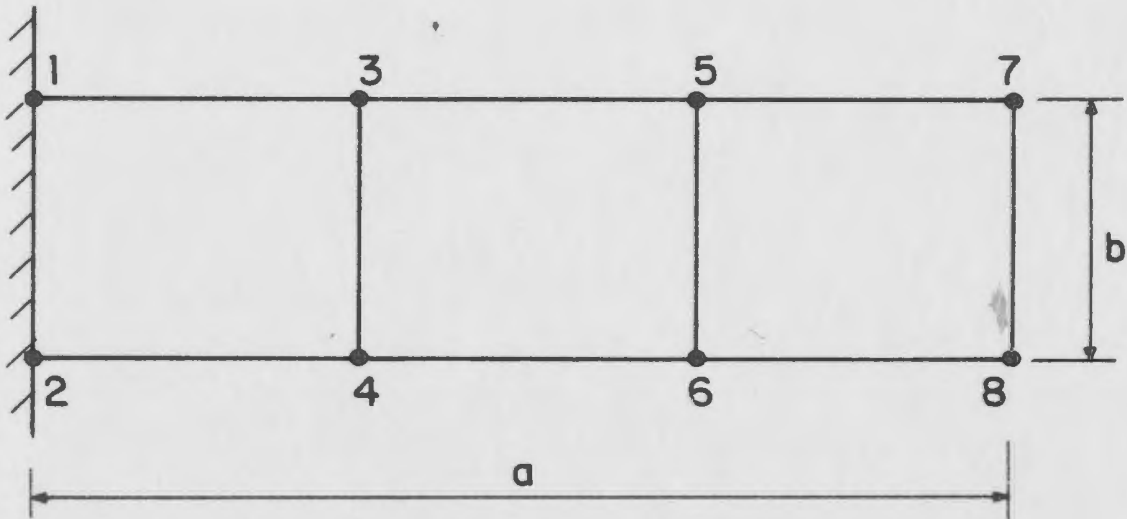


FIG. 3.11 CANTILEVER BEAM UNDER PERIODIC VERTICAL LOADING AT NODES 7 AND 8 (SMITH, 1982)

An eigenvalue analysis is first performed and the natural frequencies are shown in Table 3.1 and compared to those obtained by Smith (1982). The Rayleigh damping coefficients, $B_1 = 12.36$ and $B_2 = 1.2 \times 10^{-5}$ are obtained by specifying a modal damping ratios of 6.5% and 1.7% in first and second modes respectively. The response analysis is carried out for a few time steps and also compared in Table 3.2. The responses computed from the present programme compared reasonably well with those obtained by Smith (1982).

TABLE 3.1 - COMPARISON OF FREQUENCIES FOR
CANTILEVER BEAM

Mode Numbers	Present Analysis	Smith (1982)	Difference in Percent
1	96	102	6.25
2	431	458	6.26
3	496	526	6.05
4	903	919	1.77
5	1350	1422	5.33
6	1490	1575	5.70
7	1600	1630	1.88
8	1710	1800	5.26
9	1780	1810	1.69
10	1870	2030	8.56
11	2290	2580	12.66
12	2640	2960	12.12

TABLE 3.2 - COMPARISON OF RESPONSE QUANTITIES
FOR CANTILEVER BEAM

Time (Seconds)	Displacement		Velocity		Acceleration	
	Present Analysis	Smith (1982)	Present Analysis	Smith (1982)	Present Analysis	Smith (1982)
.005	0.483×10^{-5}	0.510×10^{-5}	0.290×10^{-2}	0.300×10^{-2}	1.160	1.200
.010	0.318×10^{-4}	0.330×10^{-4}	0.749×10^{-2}	0.776×10^{-2}	0.678	0.700
.015	0.755×10^{-4}	0.782×10^{-4}	0.955×10^{-2}	0.987×10^{-2}	0.146	0.144
.020	0.123×10^{-3}	0.127×10^{-3}	0.897×10^{-2}	0.912×10^{-2}	-0.390	-0.444
.025	0.161×10^{-3}	0.164×10^{-3}	0.573×10^{-2}	0.531×10^{-2}	-0.960	-1.077

CHAPTER IV

DYNAMIC SHAKEDOWN ANALYSIS OF OFFSHORE
FOUNDATION SUBJECTED TO CYCLIC LOADING4.1 Introduction

This chapter presents the dynamic shakedown analysis of a gravity type offshore foundation under cyclic loading conditions. Although the shakedown analysis is relatively new to the field of geotechnical engineering, its existence in literature related to structural mechanics has been well known for more than four decades.

For foundations subjected to loads varying in time in a nonproportional manner within prescribed limits, the classical limit theorems can give unsafe estimates of the collapse load, as failure can occur at loads well below the static collapse values. Shakedown theorems, which are generalizations of the limit theorems, would provide appropriate safe bounds for such complex loading programmes. For load reversal varying between prescribed limits, a foundation is said to shakedown when a state is reached such that all subsequent load applications produce only elastic

changes. In recent years, very few attempts have been made to find practical solutions to the shakedown problem related to marine foundations resting on saturated soils. As the nonlinear dynamic response analysis of an offshore foundation to wave loading can only be carried out for a very limited number of cycles due to constraints on computing costs, the shakedown analysis becomes more practical and relevant to the overall design process.

This chapter is divided into two parts. The first part introduces the basic background and concepts on shakedown related theorems following Martin (1975). The second part describes the finite element formulation of the dynamic shakedown analysis of offshore foundation and its solution as a linear programming problem.

A computer programme has been developed based on the shakedown theory and typical solutions of two dimensional boundary value problems are obtained and the validity of the computer code is checked by comparing these solutions with the analytical results.

4.2 Assumptions

The basic assumptions of classical shakedown theory are as follows (König and Maier, 1981);

- (a) The material obeys the elastic-perfectly plastic constitutive laws. The hardening (or softening) phenomenon in the shakedown process is neglected.
- (b) The plastic flow rule is governed by the associative flow theory of plasticity i.e. normality rule is applicable. The yield surface is convex in nature and the material obeys the Drucker's postulate (1950) of material stability criteria.
- (c) Effects of configuration changes do not influence the equilibrium condition of the body i.e. geometric nonlinearity is not being considered. Hence, the classical shakedown theorems are based on 'first-order' theory.
- (d) The loading process is assumed to vary slowly in time i.e. inertia and damping forces are neglected and therefore the response process is quasistatic.
- (e) The elastic properties governing the constitutive laws and the yield stresses are considered to be independent of temperature.

- (f) The phenomenon of creep is not considered.
- (g) The quantities involved in shakedown analysis can be expressed in deterministic form.

The assumption of elastic-perfectly plastic material implies that the yield surface remains unaltered over the deformation process. The stress vector can either be inside the yield surface (elastic domain) causing the plastic strain rates to be equal to zero or can touch the yield surface thus generating non-vanishing plastic strain rates (plastic domain).

The safe soil stress state, σ_{ij}^s , may be defined by the stress state corresponding to a point lying inside the yield surface whereas the admissible soil stress state, σ_{ij}^a , may be defined by the stress state corresponding to points including those on the yield surface. Here, σ_{ij} , denotes the cartesian stress tensor. The yield condition, f_y , can be written in the following forms:

$$f_y (\sigma_{ij}^s) < 0 \quad (\text{elastic}) \quad \dots (4.1a)$$

and

$$f_y (\sigma_{ij}^a) \leq 0 \quad (\text{elastic-plastic}) \quad \dots (4.1b)$$

The Drucker postulate of stability criteria states that the work of additional stresses on a closed stress path is non-negative, or that no energy can be recovered from the initially stressed element.

The postulate is presented in the form

$$\left(\sigma_{ij} - \sigma_{ij}^s \right) \dot{\epsilon}_{ij}^p > 0 \quad \dots (4.2a)$$

and

$$\left(\sigma_{ij} - \sigma_{ij}^a \right) \dot{\epsilon}_{ij}^p \geq 0 \quad \dots (4.2b)$$

where σ_{ij} = total elastic-plastic stress state of bulk solid corresponding to strain, ϵ_{ij} .

4.3 Shakedown Analysis

Shakedown analysis is the generalization of the limit analysis in a sense that it provides information whether a body will shakedown or not under a given cyclic loading programme in contradistinction to the limit analysis which provides the information whether or not flow will take place in the same body under given loads.

4.3.1 Static Shakedown

Based on Melan's theorem, when an elastic-perfectly plastic body is subjected to a load, $\hat{T}_i(\bar{x}, t)$, varying slowly between prescribed limits, it will shakedown if a time independent residual stress field, $\sigma_{ij}^R(\bar{x})$, can be found such that

$$f_y \left\{ \sigma_{ij}^E(\bar{x}, t) + \sigma_{ij}^R(\bar{x}) \right\} \leq 0 \quad \dots (4.3)$$

for every \bar{x} and t . $f_y(\sigma_{ij}) \leq 0$ is the yield condition and, $\sigma_{ij}^E(\bar{x}, t)$, is the linear elastic stress response of the body to the loading programme, $\hat{T}_i(\bar{x}, t)$. In the above expressions, σ_{ij} , \bar{x} , and t denote the cartesian stress tensor, the coordinate system and time respectively.

4.3.2 Kinematic Shakedown

According to Koiter's theorem, the continuum or the body will not shakedown i.e., it will fail ultimately by cyclic plastic deformation, if any admissible plastic strain rate cycle, $\dot{\epsilon}_{ij}^P$, can be found over some time interval, τ , such that the following inequality holds:

$$\int_0^\tau dt \left(\int_{S_2} \hat{T}_i \dot{u}_i dS + \int_V B_i \dot{u}_i dV \right) > \int_0^\tau dt \int_V Q(\dot{\epsilon}_{ij}^P) dV \quad \dots (4.4)$$

where $\hat{T}_i =$ surface tractions prescribed over,
 S_2 (Fig. 3.1)

$B_i =$ body force over the entire volume, V

$\dot{u}_i =$ admissible velocity field

and

$Q(\dot{\epsilon}_{ij}^p) =$ plastic dissipation density function
 uniquely determined by the
 kinematically admissible strain
 rate distribution, $\dot{\epsilon}_{ij}^p$

Conversely, if shakedown occurs for all admissible
 cycles of plastic strain rates and arbitrary loads
 varying slowly between prescribed limits, then it
 is possible to find a scalar multiplier, $\beta_s \geq 1$,
 such that

$$\beta_s \int_0^\tau dt \left(\int_{S_2} \hat{T}_i \dot{u}_i dS + \int_V B_i \dot{u}_i dV \right) = \int_0^\tau dt \int_V Q(\dot{\epsilon}_{ij}^p) dV$$

... (4.5)

The maximum value of, β_s , is then obviously the
 factor of safety with respect to shakedown.

4.3.3 Dynamic Shakedown

Based on Ceradini's dynamic shakedown theorem, the continuum subjected to dynamic loading will shakedown if among all the systems of initial conditions (including the self-equilibrating stress state as well as the distribution of initial displacements and velocities) there exist both a residual stress field and an elastic solution of the dynamic problem so that their sum gives stresses within the plastic limits. The elastic dynamic response has often been called as a fictitious process which is determined by imposing the arbitrary initial conditions such as displacement and velocity at some time which need not coincide with the actual ones.

In the case of a periodic dynamic loading, the possibility of shakedown is in fact determined by the forced vibration part of the response which is independent of the initial conditions of motion. The free vibration will decay in the early stage due to viscous properties of the system. Therefore, the dynamic shakedown problem may be reduced to a static case with the application of the proper amplification factor on the response.

For the sake of completeness and clarity, the proofs

of the above theorems are presented in Appendix A.

4.4 Cyclic Loading

The foundation of a gravity type offshore structure is quite often subjected to repeated or cyclic loading (Fig. 1.1'). This loading is primarily caused by wave action. A typical cyclic loading programme can be defined as follows:

$$\hat{T}_i(\bar{x}, t) = \hat{T}_i(\bar{x}, t + \omega) \quad \dots (4.6)$$

where t is any parameter which increases monotonically with real time, and, ω , is its period.

The soil stress distribution can be written as the sum of the elastic and the residual stress distributions as

$$\sigma_{ij}(\bar{x}, t) = \sigma_{ij}^E(\bar{x}, t) + \sigma_{ij}^R(\bar{x}, t) \quad \dots (4.7)$$

The elastic soil stress distribution is also a function of cyclic loading and therefore it depends on the instantaneous value of the external loads. Then, σ_{ij}^E , can be expressed as

$$\sigma_{ij}^E(\bar{x}, t) = \sigma_{ij}^E(\bar{x}, t + \omega) \quad \dots (4.8)$$

Because of the cyclic nature of the loading programme, the total soil stress distributions, $\sigma_{ij}(\bar{x}, t)$ and $\sigma_{ij}(\bar{x}, t + \omega)$, are in equilibrium with the external loadings, $\hat{T}_i(\bar{x}, t)$ and $\hat{T}_i(\bar{x}, t + \omega)$, respectively. The stress field differences are also self-equilibrating i.e. in equilibrium with zero external forces.

4.4.1 Shakedown

Shakedown is said to occur when the plastic strain rates are zero. The foundation will attain a cyclic steady state after N cycles i.e.

$$\dot{\epsilon}_{ij}^P(\bar{x}, t) = 0 \quad \text{for } t > N\omega \quad \dots (4.9)$$

and the displacements, $U_i(\bar{x}, t)$, are also cyclic. The time independent residual stress field, $\sigma_{ij}^R(\bar{x})$, is uniquely determined by the plastic strain, $\epsilon_{ij}^P(\bar{x}, t)$. The total stress is expressed from Eqn. (4.7) as

$$\sigma_{ij}(\bar{x}, t) = \sigma_{ij}^E(\bar{x}, t) + \sigma_{ij}^R(\bar{x})$$

The behaviour of the foundation can be considered in terms of the response to one cycle of loading such that $N\omega < t < (N + 1)\omega$. The elastic strain rates are given by

$$\dot{\epsilon}_{ij}^E(\bar{x}, t) = \hat{C}_{\kappa\ell} \left[\dot{\sigma}_{ij}^E(\bar{x}, t) + \dot{\sigma}_{ij}^R(\bar{x}, t) \right] \dots (4.10)$$

where, $\hat{C}_{\kappa\ell}$, represents the coefficients of the elasticity matrix.

The change in elastic strain field over a cycle is zero and therefore

$$\Delta\epsilon_{ij}^E(\bar{x}) = 0 \dots (4.11)$$

However, the change in the total strains

$$\Delta\epsilon_{ij}(\bar{x}) = \epsilon_{ij}(\bar{x}, (N+1)\omega) - \epsilon_{ij}(\bar{x}, N\omega) \dots (4.12)$$

must be kinematically admissible, since it represents the difference between two kinematically admissible fields. The change in the plastic strains over a cycle is defined here as

$$\Delta\epsilon_{ij}^P(\bar{x}) = \int_{N\omega}^{(N+1)\omega} \dot{\epsilon}_{ij}^P(\bar{x}, t) dt \dots (4.13)$$

$\Delta\epsilon_{ij}^P(\bar{x})$ will also be kinematically admissible when the foundation reaches a cyclic steady state which determines the condition for shakedown.

4.4.2 Alternating Plasticity

For the situation when, $\Delta \epsilon_{ij}^P(\bar{x}) = 0$, it is implied that the plastic strains, $\epsilon_{ij}^P(\bar{x}, t)$, and the displacements are cyclic in nature as well as their rates. This condition leads to the situation where the elements in the system undergo plastic strains which are alternating in nature i.e. change in sign with each cycle.

4.4.3 Incremental Collapse

When, $\Delta \epsilon_{ij}^P(\bar{x}, t) \neq 0$, the foundation undergoes incremental changes in displacement, $\Delta \dot{u}_i$, over each cycle. The work done by the external loading programme in each cycle may be written as

$$\int_{N\omega}^{(N+1)\omega} \hat{T}_i \dot{u}_i dt = \int_{N\omega}^{(N+1)\omega} dt \left(\int_V \sigma_{ij} \dot{\epsilon}_{ij}^E dV + \int_V \sigma_{ij} \dot{\epsilon}_{ij}^P dV \right) \dots (4.14)$$

Using Eqn. (4.11), Eqn. (4.14) reduces to

$$= \int_{N\omega}^{(N+1)\omega} dt \left(\int_V \sigma_{ij} \dot{\epsilon}_{ij}^P dV \right) > 0 \dots (4.15)$$

Thus it can be seen the foundation will undergo increasing deformation with each cycle and if the

cycling loading continues for a long period of time, the foundation will eventually fail as a result of unbounded deformation. This type of behaviour is often termed as progressive deformation or incremental collapse.

From the offshore' foundation design point of view, a foundation subjected to a combination of cyclic loading (e.g. vertical, horizontal etc.), in which shakedown does not occur represents a situation where failure is almost certain.

4.5 Nonlinear Programming Approach to Shakedown Analysis

Consider a body subjected to cyclic loads, $\hat{T}(\bar{x}, t)$ with an elastic solution, $\sigma_{ij}^E(\bar{x}, t)$. Due to the linearity of the elastic solution, if the loads are magnified by β_s times i.e. $\beta_s \hat{T}_i(\bar{x}, t)$, (where β_s is a scalar multiplier and $\beta_s > 1$), the elastic stress field will be, $\beta_s \sigma_{ij}^E(\bar{x}, t)$. From Melan's theorem, if shakedown occurs for a multiplier, $\beta_s > 1$, there will be a shakedown residual stress field, $\sigma_{ij}^R(\bar{x})$, such that

$$f_y \{ \beta_s \sigma_{ij}^E(\bar{x}, t) + \sigma_{ij}^R(\bar{x}) \} \leq 0 \quad \dots (4.16)$$

The vector, $\{ \beta_s \sigma_{ij}^E(\bar{x}, t) + \sigma_{ij}^R(\bar{x}) \}$, must not

exceed the yield limit at any point in the body during the loading cycle.

As the yield function, $f_y (\sigma_{ij}) \leq 0$, is normally a convex quadratic function in stress space, the evaluation of the shakedown load factor, β_s , based on Melan's theorem leads to the following programming problem:

$$\beta = \text{Maximize } \beta_s$$

$$\text{subject to } f_y \{ \beta_s \sigma_{ij}^E (\bar{x}, t) + \sigma_{ij}^R (\bar{x}) \} \leq 0 \quad \dots (4.17)$$

The determination of β thus becomes a nonlinear mathematical programming problem.

In order to apply Koiter's theorem for the evaluation of the shakedown load factor, β , the external work is equated with internal work for any admissible plastic strain rate cycle, $\dot{\epsilon}_{ij}^P (\bar{x}, t)$. Therefore, β , can be obtained by equating the external work to the internal energy.

From Eqn. (4.5) one obtains

$$\beta = \frac{\int_0^\tau \int_V Q (\dot{\epsilon}_{ij}^P) dV}{\int_0^\tau \left(\int_{S_2} \hat{T}_i (\bar{x}, t) \dot{u}_i (\bar{x}, t) dS + \int_V B_i \dot{u}_i (\bar{x}, t) dV \right)} \quad \dots (4.18)$$

It is obvious then from Koiter's theorem that β is the smallest multiplier obtained by equating external and internal work. Thus any value obtained from Koiter's theorem provides an upper bound on the shakedown load factor whereas the value obtained from Melan's theorem is lower bound.

4.6 Evaluation of Shakedown Load Factor

The determination of the shakedown load factor often requires the numerical solution of the optimization problems given by Eqn. (4.17) or Eqn. (4.18).

Therefore, the constraints given by Eqn. (4.17) or Eqn. (4.18) must be solved by suitably discretizing the loading programme, stress field and the yield condition.

4.6.1 Discretization of the Loading Programme

The loading domain is represented in load space by, $F(\hat{T}_i) = 0$, and shown in Fig. 4.1. It can also be represented as an unit loading domain, $F_o(\hat{T}_i) = 0$, multiplied by the appropriate load factor, β_s . The actual loading history, $\hat{T}_i(\bar{x}, t)$, can pass through any point inside the loading domain, $F(\hat{T}_i) = 0$, as a certain function of time, t . As the foundation is subjected to a cyclic loading programme which passes

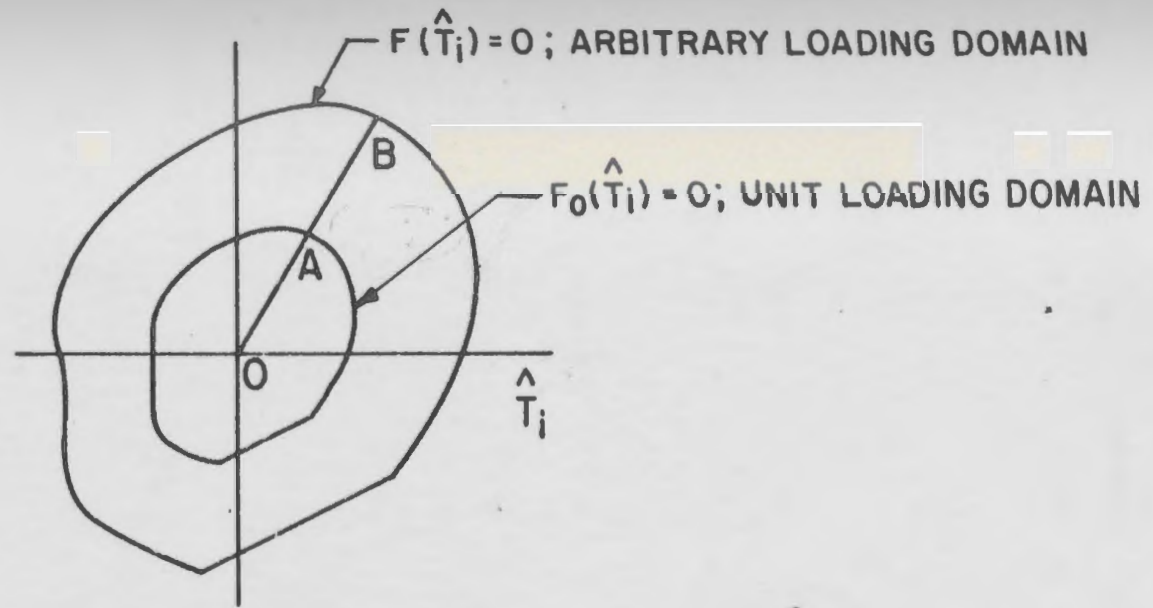


FIG. 4.1 GENERALIZED LOADING DOMAIN ($\frac{OB}{OA} = \beta$; HO, 1972)

beyond the initial yield surface, the yield surface will start moving and form a subsequent yield surface which changes with time. Shakedown will occur if a yield surface can be set up such that it contains the loading cycle, $\hat{T}_i(\bar{x}, t)$, which means the response thereafter will be purely elastic. On the other hand, if shakedown does not occur, the yield surface will continue to translate each time the load cycle is repeated with increased plastic deformation leading to incremental collapse.

The time dependent variable repeated loading function, $\hat{T}_i(\bar{x}, t)$, can be expressed as

$$\hat{T}_i(\bar{x}, t) = \sum_{s=1}^n \gamma_s(t) \bar{T}_s(\bar{x}) \quad \dots (4.19)$$

where $\bar{x} \in S_2$ (Fig. 3.1)

$\gamma_s(t)$ = s-th load parameter

$\bar{T}_s(\bar{x})$ = constant surface traction (reference) in the s-th unit load mode

The simplest load domain Ω in n-dimensional space is shown in Fig. 4.2 and it is assumed that the load factor, $\gamma_s(t)$, can vary within the load space. The load factor domain is also assumed to be convex.

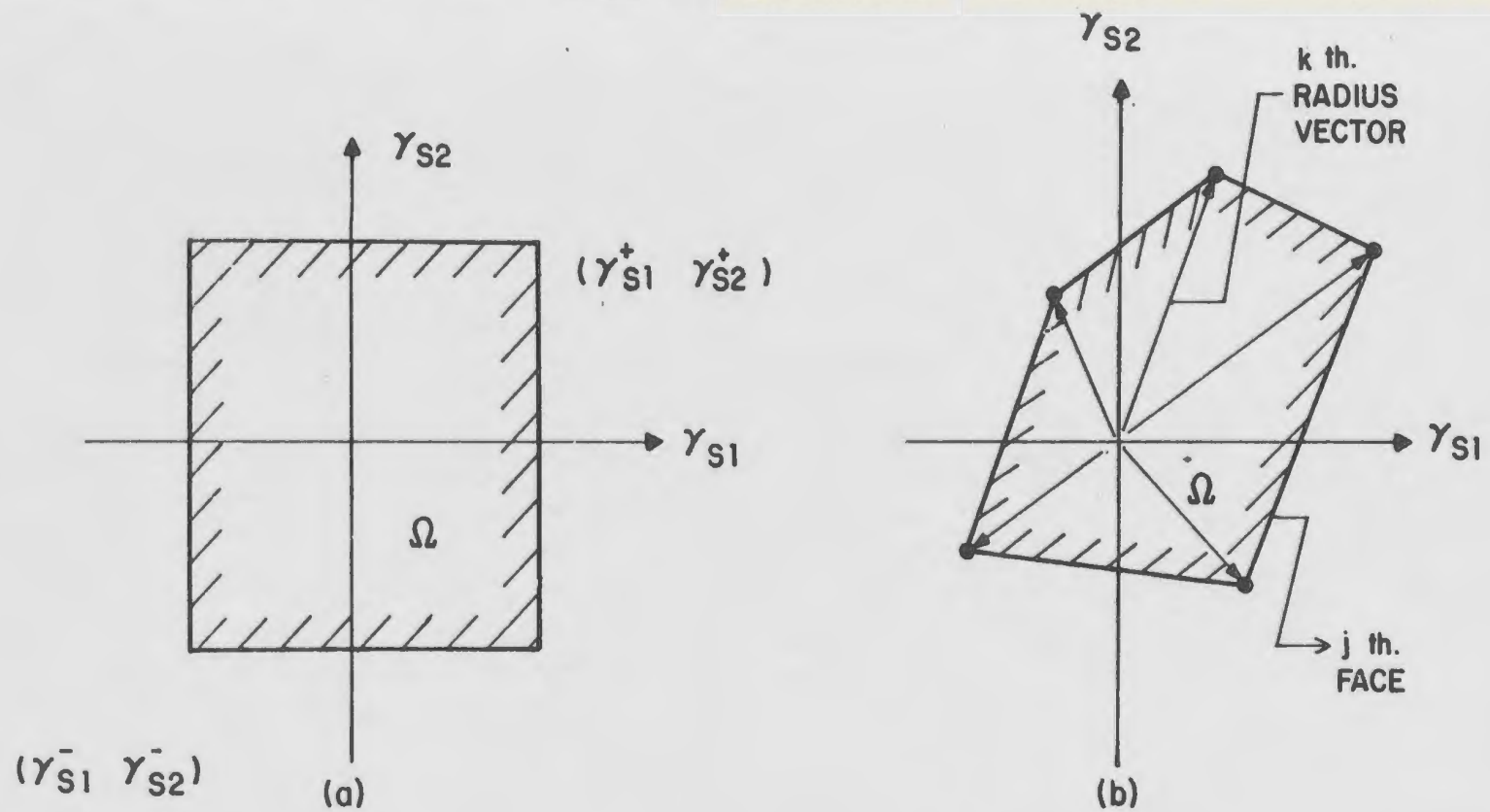


FIG. 4.2 DIFFERENT LOADING DOMAINS FOR SHAKEDOWN ANALYSIS (a) RECTANGULAR LOADING DOMAIN (b) ARBITRARY LOADING DOMAIN (BORKOWSKI AND KLEIBER, 1980)

Denoting the corners of the load domain by $\gamma_{s1}, \dots, \gamma_{sn}$, and the unique respective elastic stress responses corresponding to these load factors as $\sigma_{ij}^{1E}(\bar{x}), \dots, \sigma_{ij}^{nE}(\bar{x})$, it can be shown that a given system shakes down to a certain cyclic loading process which contains all the vertices of a given load domain Ω , (e.g. $\gamma_{s1}, \dots, \gamma_{sn}$), then it shakes down to any load path contained within Ω . Therefore, the mathematical problem of Eqn. (4.17) is expressed as follows:

$$\beta = \text{Max } \beta_s \quad \dots (4.20)$$

subject to

$$f_y \left\{ \beta_s \sigma_{ij}^{kE}(\bar{x}) + \sigma_{ij}^R(\bar{x}) \right\} \leq 0 \text{ in } V$$

where $k = 1, 2, \dots, n$

4.6.2 Discretization of the Stress Field

In order to compute, $\sigma_{ij}^{kE}(\bar{x}, t)$, in the continuum formulation, it is necessary to discretize the stress field by using finite difference or finite element method. It is also necessary to express the residual stress field, $\sigma_{ij}^R(\bar{x})$, in terms of a discrete number of parameters. Finally, the yield

condition is applied at discrete number of points in the body defined here as 'check points'.

4.6.3 Piecewise Linearization of the Yield Surface

The evaluation of shakedown load factor, β , from Eqn. (4.20) may be carried out as a nonlinear optimization problem by satisfying the yield condition at the selected 'check points' in the discretized body. However, this may lead to computational difficulty for large class of continuum problems. Therefore, the piecewise linearization of the yield function is used here in order to transfer a nonlinear mathematical programming problem to a linear programming one. The piecewise linearization of the convex yield function requires that the yield domain be a polyhedron in the stress space and represented by a finite number of yield planes. This means that instead of satisfying nonlinear constraints at selected 'check points' in the discretized body, it is sufficient to satisfy a large number of linear inequalities at those 'check points'. Therefore, the shakedown load factor, β_s , can be obtained by solving a linear programming problem (Maier, 1969).

Assuming, $f_y(\sigma_{ij}, \epsilon_{ij}^p) \leq 0$, a differentiable yield function such that, $f_y(\sigma_{ij}) \leq 0$ denotes the current

elastic domain in the, σ_{ij} , space, one may write

$$\dot{f}_y = \frac{\partial f_y}{\partial \sigma_{ij}} \dot{\sigma}_{ij} + \frac{\partial f_y}{\partial \varepsilon_{ij}^p} \dot{\varepsilon}_{ij}^p \quad \dots (4.21)$$

The plastic strain rate satisfying the normality condition is defined by

$$\dot{\varepsilon}_{ij}^p = \dot{\Delta} \frac{\partial f_y}{\partial \sigma_{ij}} \quad \dots (4.22)$$

where Δ = plastic multiplier.

The plastic flow rule at the yield limit, $f_y = 0$, requires $\dot{\Delta} \geq 0$, $f_y \leq 0$, $\dot{f}_y \dot{\Delta} = 0$ and $f_y \dot{\Delta} = 0$. The last condition is often termed as a complementary condition, which implies that plastic yielding (loading process) is active only when the current stress, σ_{ij} , is in contact with yield surface, $f_y = 0$, and loss of contact of the stress point with the yield surface ($\dot{f}_y \leq 0$, unloading process) are events which are mutually exclusive. Writing the above equations as follows, one obtains

$$\dot{f}_y = \tilde{N}^T \dot{\sigma}_{ij} - H \dot{\Delta} \quad \dots (4.23a)$$

where

$$\tilde{N} = \frac{\partial f_y}{\partial \sigma_{ij}} \quad \dots (4.23b)$$

and

$$H = - \frac{\partial f_y}{\partial \epsilon_{ij}^p} \cdot \frac{\partial f_y}{\partial \sigma_{ij}} \quad \dots (4.23c)$$

The scalar H is defined here as the 'hardening modulus', where $H > 0$ represents work hardening, $H = 0$ elastic-perfectly plastic and $H < 0$ softening behaviour. Of course, $H < 0$, is not permitted because of the Drucker's stability postulate and associative nature of the flow rule (normality condition).

Linearizing the continuously differentiable yield function, $f_y(\sigma_{ij}, \epsilon_{ij}^p) \leq 0$, into a suitable number of yield planes, R , one can write in matrix form the following

$$\left\{ \dot{f}_{yi} \right\} = \left[\tilde{N}_i \right]^T \left\{ \dot{\sigma}_{rs} \right\} - \left[H_i \right] \left\{ \dot{\Delta}_i \right\} - \left\{ K_{oi} \right\} \leq 0 \quad \dots (4.24a)$$

where

$$\left[\tilde{N}_i \right]^T = \frac{\partial f_{yi}}{\partial \sigma_{rs}} \quad \dots (4.24b)$$

$$\left[H_i \right] = - \frac{\partial f_{yi}}{\partial \epsilon_{rs}^p} \cdot \frac{\partial f_{yi}}{\partial \sigma_{rs}} \quad \dots (4.24c)$$

$$i = (1, 2, \dots R)$$

and

$\{K_{oi}\}$ = each element in the vector defines the distance of each yield plane, $f_{yi} = 0$, from the origin (Fig. 4.3)

Therefore, Eqn. (4.22) is rewritten as

$$\dot{\epsilon}_{rs}^p = \sum_{i=1}^R \frac{\partial f_{yi}}{\partial \sigma_{rs}} \dot{\Delta}_i \quad \dots (4.25)$$

The graphical interpretation of Eqn. (4.24) is shown in Fig. 4.3.

Representing the yield function, f_{yi} , for a typical 'check point', j , within the body for ideal plasticity ($H = 0$) one can write

$$\{f_{yi}^j\} = \left[\tilde{N}_i^j \right]^T \{ \dot{\sigma}_{rs}^j \} - \{K_{oi}^j\} \leq 0 \quad \dots (4.26)$$

Eqn. (4.26) describes the inequality that represents the elastic polyhedron at 'check point' j , in which each of its $i = 1, 2, \dots R$ yield planes is identified with an unit normal vector in, $\left[\tilde{N}_i^j \right]$, and a element in vector, $\{K_{oi}^j\}$, which represents the distance of each yield plane from the origin (Fig. 4.3).

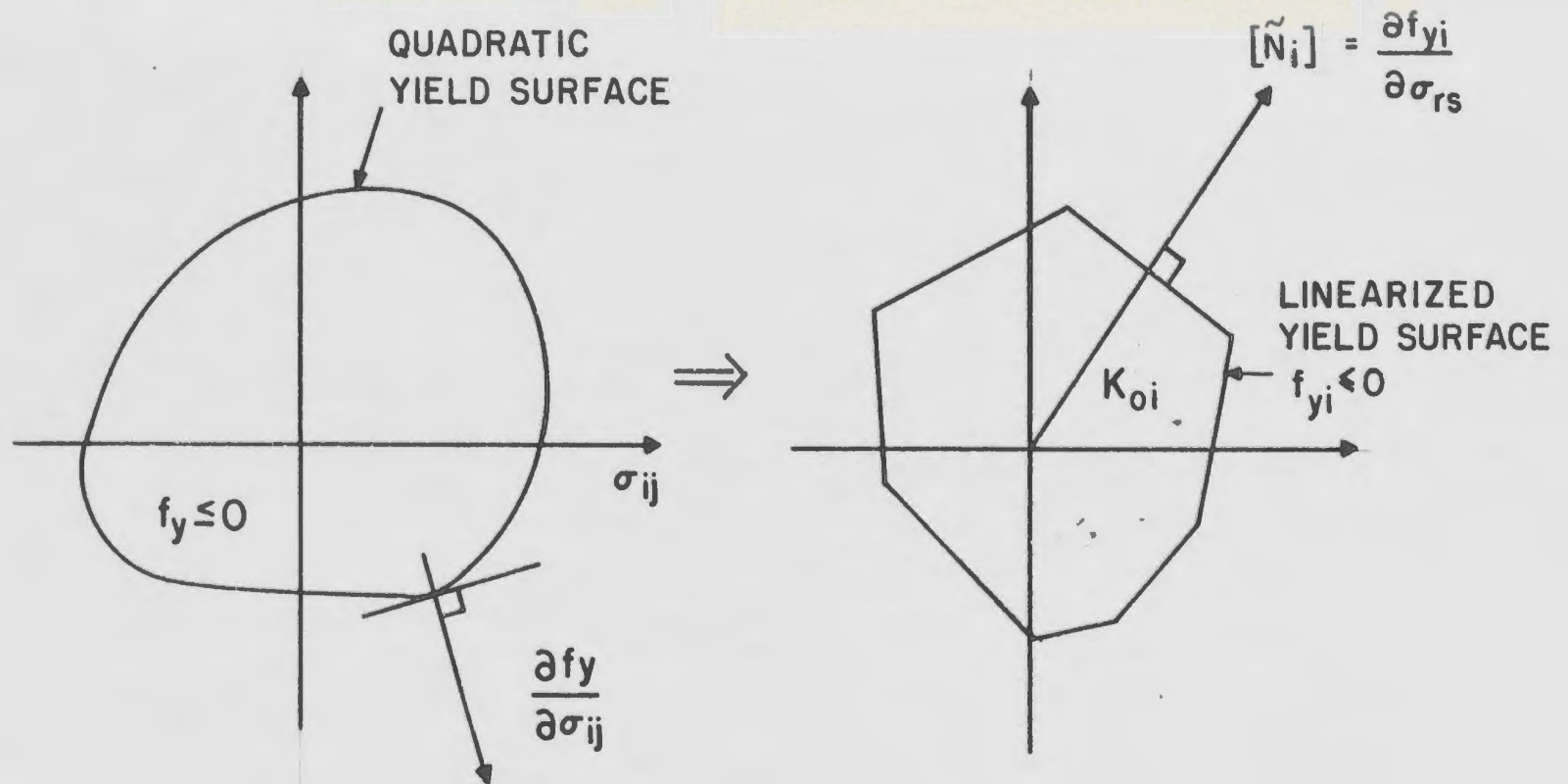


FIG. 4.3 PIECEWISE LINEARIZED YIELD SURFACE (MAIER, 1976)

4.7 Finite Element Formulation for Dynamic Shakedown Analysis

This section describes the application of the finite element method to determine the shakedown load for the foundation of a gravity type offshore structure subjected to dynamic cyclic loading. For the sake of numerical computation, the foundation is discretized into an assemblage of four noded isoparametric elements following Haldar et al (1982a and 1982b). In the present finite element model, displacement compatibility across the boundary between two adjacent elements is satisfied whereas the equilibrium condition will be satisfied only in a global sense.

The foundation is discretized with M number of isoparametric finite elements having N^0 number of unconstrained displacements and a finite number of subvolumes, V_j , ($j = 1, 2, \dots M$).

4.7.1 Evaluation of Internal Work

The plastic energy dissipation rate, $Q (\dot{\epsilon}_{rs}^{pj})$, associated with a subvolume, V_j , can now be expressed as

$$Q (\dot{\epsilon}_{rs}^{pj}) = \int_{V_j} \left\{ \dot{\epsilon}_{rs}^{pj} \right\}^T \left\{ \sigma_{rs}^j \right\} dV \quad \dots (4.27)$$

and substituting Eqn. (4.25) in Eqn. (4.27) results in

$$\begin{aligned} Q (\dot{\epsilon}_{rs}^{pj}) &= \int_{V_j} \sum_{i=1}^R \frac{\partial f_{yi}}{\partial \sigma_{rs}} \dot{\Delta}_i^j \left\{ \sigma_{rs}^j \right\} dV \\ &= \left\{ \dot{\Delta}_{\sim i}^j \right\}^T \left[\tilde{N}_i^j \right]^T \left\{ \sigma_{rs}^j \right\} \quad \dots (4.28) \end{aligned}$$

where $\left\{ \dot{\Delta}_{\sim i}^j \right\}^T = \int_{V_j} (\dot{\Delta}_1^j \dots \dot{\Delta}_R^j) dV$

and $\left[\tilde{N} \right] =$ matrix whose columns contain $\frac{\partial f_{yi}}{\partial \sigma_{rs}}$

Therefore the total plastic energy dissipation rate for the entire foundation is obtained by assembling $Q (\dot{\epsilon}_{rs}^{pj})$ over all elements as

$$\begin{aligned} \dot{Q} &= \sum_{j=1}^M Q (\dot{\epsilon}_{rs}^{pj}) \\ &= \sum_{j=1}^M \left\{ \dot{\Delta}_{\sim i}^j \right\}^T \left[\tilde{N}_i^j \right]^T \left\{ \sigma_{rs}^j \right\} \\ &= \sum_{j=1}^M \left\{ \dot{\Delta}_{\sim i}^j \right\}^T \left\{ f_{yi}^j + K_{oi}^j \right\} \\ &= \left\{ K_o \right\}^T \left\{ \dot{\Delta}_{\sim i} \right\} \quad \dots (4.29) \end{aligned}$$

where $\left\{ K_o \right\}$ and $\left\{ \dot{\Delta}_{\sim i} \right\}$ collect all the components of $\left\{ K_{oi}^j \right\}$ and $\left\{ \dot{\Delta}_{\sim i}^j \right\}$ for all elements ($j = 1, 2, \dots, M$), and

every yielding mode, ($i = 1, 2, \dots, R$). Correspondingly, the vector, $\{\hat{f}_y\}$, collects all $\{f_{yi}^j\}$.

The internal work is then evaluated by integrating Eqn. (4.29) over one cycle, τ ,

$$\begin{aligned} W_{int} &= \int_0^{\tau} \dot{Q} \, d\tau = \int_0^{\tau} \{K_o\}^T \left\{ \begin{array}{c} \dot{\Delta} \\ \approx \end{array} \right\} d\tau \\ &= \{K_o\}^T \int_0^{\tau} \left\{ \begin{array}{c} \dot{\Delta} \\ \approx \end{array} \right\} d\tau \\ &= \{K_o\}^T \left\{ \begin{array}{c} \Delta \\ \approx \end{array} \right\} \quad \dots \quad (4.30) \end{aligned}$$

4.7.2 Kinematically Admissible Plastic Strain

From Eqn (3.381), the bulk load vector, $\{p_B\}$, for an element, j , can be written in the following form:

$$\{p_B^j\} = \int_V \phi_m^T \left\{ \begin{array}{c} X_B \\ Y_B \end{array} \right\} dV + \int_{S_2} \phi_m^T \left\{ \begin{array}{c} \hat{T} \end{array} \right\} dS$$

Assembling over all elements, and using Eqn. (3.25), it can now be written as

$$[G_o]^T \{\sigma\} = \{P_B\} \quad \dots \quad (4.31)$$

where $[G_o]^T$ collects all the integrals, $\sum \int_{V_j}^M [\psi^j] \left\{ \begin{array}{c} \psi^j \\ \approx \end{array} \right\}^T dV$,

$\{P_B\}$ collects all bulk load vectors, $\sum \{p_B^j\}$

for the entire assemblage and therefore, $\{\sigma\}$, represents the collection of all stress components. If no loads are applied to the body, only self equilibrating residual stress field will be present and equilibrium is provided by

$$[G_o]^T \{\sigma\} = 0 \quad \dots (4.32)$$

From Eqn. (4.5), for an element j , the external work is equated with the internal energy dissipated during one cycle, τ , and therefore

$$\int_0^\tau \{p_B^j\}^T \{\dot{U}^j\} dt = \int_0^\tau \left(\int_{V_j} \{\sigma^j\}^T \{\dot{\epsilon}^{pj}\} dV_j \right) dt \quad \dots (4.33)$$

Substituting Eqn. (3.25) in Eqn. (4.33) for element, j

$$\int_0^\tau \int_{V_j} \{\sigma^j\}^T [\underline{\psi}^j] \{\dot{U}^j\} dV_j = \int_0^\tau \left(\int_{V_j} \{\sigma^j\}^T \{\dot{\epsilon}^{pj}\} dV_j \right)$$

and using Eqn. (4.28)

$$\left(\int_{V_j} [\underline{\psi}^j] dV_j \right) \{\underline{U}^j\} = [\tilde{N}^j] \{\underline{\Delta}^j\} \quad \dots (4.34)$$

Assembling over all elements $j = 1, 2, \dots, M$, one can write

$$[G_o] \{\underline{U}\} = [\tilde{N}] \{\underline{\Delta}\} \quad \dots (4.35)$$

where $[G_0]$ is known as the assembled compatibility matrix of the entire foundation system. $[\tilde{N}] = (\text{diag. } \tilde{N}^1 \tilde{N}^2 \dots \tilde{N}^M)$ represents the matrix containing gradient submatrices $[\tilde{N}^j]$ along its main diagonal. Each of the block diagonal matrices collects the column vectors $[\tilde{N}_i^j]$ of all the yield planes of each element as

$$[\tilde{N}_i^j] = [\tilde{N}_1^j, \tilde{N}_2^j \dots \tilde{N}_R^j] \quad \dots (4.36)$$

Correspondingly, the vector, $\{\underline{U}\}$ represents the nodal displacements of the discretized system.

4.7.3 Evaluation of External Work

Using the virtual work principle and assembling over all elements, the total external work during one cycle can be expressed as

$$W_{\text{ext}} = \sum_{j=1}^M \int_0^{\tau} \left\{ \dot{U}^j \right\}^T \left\{ p_B^j \right\} d\tau = \sum_{j=1}^M \int_0^{\tau} d\tau \int_{V_j} \left\{ \sigma^{Ej} \right\}^T \left\{ \dot{\epsilon}^{pj} \right\} dV \quad \dots (4.37)$$

Assume that $\dot{\epsilon}^p$ is a vector which has a fixed direction in strain space. Let $\{\sigma^E\}_{\text{max}}$ indicate the value of $\{\sigma^E\}$ which has the largest component

in the direction of $\{\dot{\epsilon}^p\}$. Then the following inequality holds

$$\left\{\sigma^E\right\}_{\max}^T \left\{\dot{\epsilon}^p\right\} \geq \left\{\sigma^E\right\}^T \left\{\dot{\epsilon}^p\right\} \quad \dots (4.38)$$

Substituting Eqn. (4.38) into Eqn. (4.37) and using Eqn. (4.28) gives'

$$W_{\text{ext}} = \int_{\tau} d\tau \sum_{j=1}^M \left\{\sigma^{Ej}\right\}_{\max}^T \left[\tilde{N}^j\right] \left\{\dot{\Delta}^j\right\} \quad \dots (4.39)$$

Eqn. (4.39) is rewritten in the following form as

$$W_{\text{ext}} = \int_{\tau} d\tau \sum_{j=1}^M \left\{\dot{\Delta}^j\right\}^T \left[\tilde{N}^j\right]^T \left\{\sigma^{Ej}\right\}_{\max} \quad \dots (4.39a)$$

Using Eqn. (4.30) and Eqn. (4.39a), Eqn. (4.18) can be rewritten as a programming problem:

$$\beta = \text{minimize } \left\{K_o\right\}^T \left\{\dot{\Delta}\right\}$$

subject to

$$\int_{\tau} d\tau \sum_{j=1}^M \left\{\dot{\Delta}^j\right\}^T \left[\tilde{N}^j\right]^T \left\{\sigma^{Ej}\right\}_{\max} = 1$$

where $\dot{\Delta} \geq 0$

and $\dot{\epsilon}^{pj}$ is compatible ... (4.40)

As described in Chapter 3, the soil beneath the gravity foundation will be in a state of undrained condition during the event of a storm which may typically last for several hours and therefore, the dynamic equations of motion for the discretized system under undrained state can be written in assembled form following Eqn. (3.48) as

$$[M^*] \left\{ \ddot{\underline{U}} \right\} + [C^*] \left\{ \dot{\underline{U}} \right\} + [K^*] \left\{ \underline{U} \right\} = \left\{ P_B \right\}$$

Here, the matrix $[K^*]$, represents the undrained stiffness matrix of the entire assemblage defined in Eqn. (3.50) and is expressed as

$$[K^*] = \sum^M \left(\int_V \psi_{11}^T D_{sd} \psi_{11} dV + \int_V \alpha^2 M_O \psi_{22}^T \psi_{22} dV \right)$$

4.8 Condition For Dynamic Shakedown

The necessary condition of shakedown for a discretized system under dynamical situation can be stated as follows:

"If the system shakes down, then some constant plastic multiplier, Δ , some initial conditions $\underline{U}^*(\bar{x}, t^*)$ $\dot{\underline{U}}^*(\bar{x}, t^*)$ and a time t^* , exist such that the yield condition can be satisfied at any time $t \geq t^*$ as follows:

$$\left\{ \begin{matrix} f_y^* \\ \end{matrix} \right\} = \left[\tilde{N} \right]^T \left\{ \begin{matrix} \sigma^* (\bar{x}, t) \\ \end{matrix} \right\} - \left\{ K_o \right\} \leq 0 \quad \dots (4.41)$$

The asterisk sign represents the fictitious process with suitably adjusted initial conditions so that the response $\sigma^*(t)$, becomes periodic (cyclic) after time $t \geq t^*$.

Substituting the value of, $\sigma^*(\bar{x}, t)$, from Eqn. (4.7) into Eqn. (4.41), one obtains

$$\left\{ \begin{matrix} f_y^* \\ \end{matrix} \right\} = \left[\tilde{N} \right]^T \left\{ \begin{matrix} \sigma^{*E}(\bar{x}, t) + \sigma^R(\bar{x}) \\ \end{matrix} \right\} - \left\{ K_o \right\} < 0 \quad \dots (4.42)$$

where, $\sigma^R(\bar{x})$, defines the time-independent residual stress field.

$\left\{ \begin{matrix} \sigma^{*E}(\bar{x}, t) \\ \end{matrix} \right\}$ is defined here as the fictitious linear elastic dynamic stress response of the undrained soil medium to the given cyclic loading, $\left\{ \begin{matrix} P_B(t) \\ \end{matrix} \right\}$, after a finite number of cycles. The stress, $\sigma^{*E}(\bar{x}, t)$, is based on initial conditions of displacement, \underline{U}^* , and velocity $\dot{\underline{U}}^*$, which need not coincide with the normally assumed zero initial conditions $\underline{U}(\bar{x}, 0) = 0.0$ and $\dot{\underline{U}}(\bar{x}, 0) = 0.0$.

The fictitious elastodynamic stress response of each element, j , defined as, $\sigma^{*Ej}(t, \underline{U}^*, \dot{\underline{U}}^*)$, is

projected on the outward normal vector, $[\tilde{N}_i^j]$, of each yield plane, i , (Fig. 4.4). Maximizing these projections with respect to time, $t \geq t^*$, gives

$$\tilde{M}_i^j = \max_{t \geq t^*} [\tilde{N}_i^j]^T \cdot \sigma^{*Ej}(t, \underline{U}^*, \dot{\underline{U}}^*) \quad \dots (4.43)$$

where

$$\{\tilde{M}_i^j\}^T = [\tilde{M}_1^j, \tilde{M}_2^j \dots \tilde{M}_R^j]_{\max}^T$$

and assembling all these \tilde{M}_i^j vectors in one place, one obtains

$$\{\tilde{M}\} = \{\tilde{M}_i^j\} \quad \dots (4.44)$$

where $i = 1, 2, \dots R$

and $j = 1, 2, \dots M$

Based on the elastic dynamic response computed from Eqn. (3.48), using initial conditions e.g. displacement $\underline{U}^* = \underline{\bar{U}}^*$ and velocity $\dot{\underline{U}}^* = \dot{\underline{\bar{U}}}^*$, the \tilde{M} -vector is evaluated from Eqn. (4.44). The maximum value of, β , of the programming problem is then expressed as

$$\beta = \max \beta_s$$

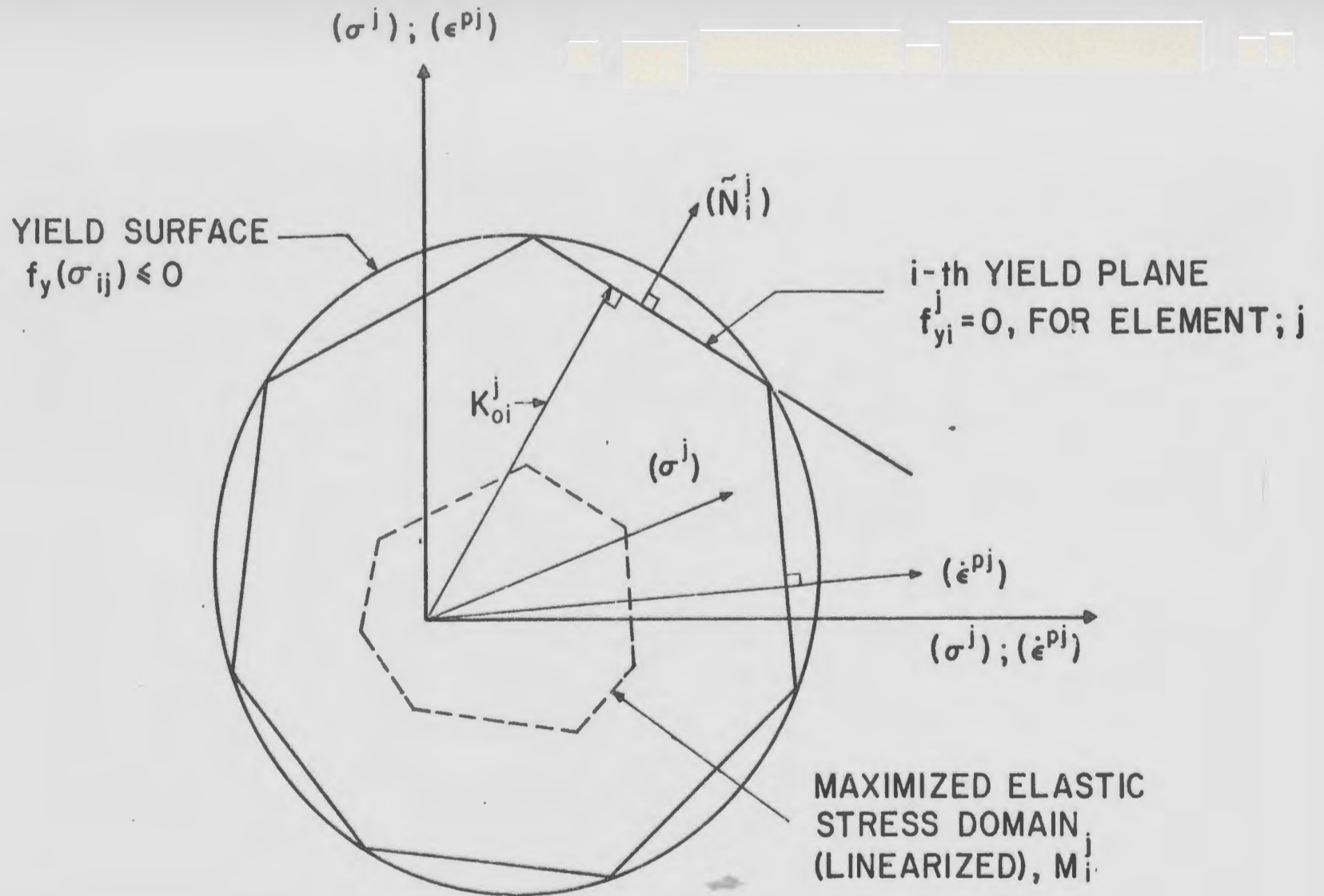


FIG. 4.4 GRAPHICAL REPRESENTATION OF LINEARIZED ELASTIC STRESS DOMAIN, MAXIMIZED WITH RESPECT TO TIME, FOR A TYPICAL ELEMENT, j .

subject to

$$\beta_s \left\{ \tilde{M} \right\} + \left[\tilde{N} \right]^T \left\{ \sigma^R \right\} \leq \left\{ K_0 \right\} \quad \dots (4.45)$$

where β_s is a common multiplier to elastic dynamic (or static) stress response and also a function of t^* , $\underline{\bar{U}}^*$, $\underline{\dot{\bar{U}}}^*$ through \tilde{M} . Assume $\underline{\bar{U}}^*$ and $\underline{\dot{\bar{U}}}^*$ as the particular initial conditions such as displacement and velocity which make the fictitious elastic response process to be periodic with respect to the external cyclic loading. For a general loading history, it may be very difficult to obtain an optimum value of β as a function of Δ , $\underline{\bar{U}}^*$ and $\underline{\dot{\bar{U}}}^*$. However, the optimization problem can greatly be simplified if the external loading process is periodic. In this case, the free vibration part of the response will decay after a finite time i.e. $t > N\omega$ due to damping and therefore the periodic response is described as

$$\sigma^{*E}(\bar{x}, t) = \sigma^{*E}(\bar{x}, t + \omega) \quad \dots (4.45a)$$

$\underline{\bar{U}}^*(t)$, $\underline{\dot{\bar{U}}}^*(t)$ represent the displacement and velocity at time $t = N\omega$, defined here as fictitious initial conditions, which make the response to be periodic i.e. independent of the free vibration part. By imposing these initial conditions at $t = 0$,

the conditions for periodicity as expressed by Eqn. (4.46) can automatically be verified for $t > 0$.

4.9 Linear Programming Formulation

4.9.1 Primal Problem

The scalar multiplier factor, β_s , is called statically admissible if Eqn. (4.45) is satisfied for some σ^R . The search for, β , is a solution of the following linear programming problem

$$\beta = \max_{\beta_s, \sigma^R} \beta_s \quad \dots (4.46)$$

subject to

$$1) \quad [G_o]^T \{ \sigma^R \} = 0$$

$$2) \quad \beta_s \left\{ \tilde{M} \right\} + [\tilde{N}]^T \{ \sigma^R \} \leq \left\{ K_o \right\}$$

$$\text{and } 3) \quad \beta_s > 0$$

which can be written in the following 'tableau' form

$$\beta = \text{Max} \begin{array}{c|c|c} \beta_s & \{\sigma^R\} & \\ \hline 1 & \{0\}^T & \\ \hline 0 & [G_o]^T & \{0\} = \\ \hline \{\tilde{M}\} & [\tilde{N}]^T & \{K_o\} \leq \end{array} \dots (4.47)$$

In order to ensure the nonnegativity of the variable $\{\sigma^R\}$, one can write

$$\{\sigma^R\} = \{\sigma^R\}^+ - \{\sigma^R\}^- \dots (4.48)$$

with $\{\sigma^R\}^+ \geq 0$ and $\{\sigma^R\}^- \geq 0$

Then the modified 'tableau' becomes

$$\beta = \text{Max} \begin{array}{c|c|c|c} \beta_s & \{\sigma^R\}^+ & \{\sigma^R\}^- & \\ \hline 1 & \{0\}^T & \{0\}^T & \\ \hline 0 & [G_o]^T & -[G_o]^T & \{0\} = \\ \hline \{\tilde{M}\} & [\tilde{N}]^T & -[\tilde{N}]^T & \{K_o\} \leq \end{array} \dots (4.49)$$

4.9.2 Dual Problem

In the linear programming formulation of an optimization problem, any maximization problem can be related to its counterpart problem such as minimization having the same vectors and matrices and connected with the former by some duality properties.

The matrix formulation of Eqn. (4.40) is furnished through Eqn. (4.30) for internal work, Eqn. (4.35) for the condition of kinematically admissible plastic strain rate cycle and Eqn. (4.39a) for external work. Therefore, the dual programme is written in the following matrix form:

$$\beta = \min_{\underline{U}, \underline{\Delta}} \left\{ K_0 \right\}^T \left\{ \underline{\Delta} \right\} \quad \dots (4.50)$$

subject to

$$a) \left[G_0 \right] \left\{ \underline{U} \right\} - \left[\tilde{N} \right] \left\{ \underline{\Delta} \right\} = 0$$

$$b) \left\{ \tilde{M} \right\}^T \left\{ \underline{\Delta} \right\} = 1$$

In order to ensure the nonnegativity of the variable, $\left\{ \underline{U} \right\}$, one can write

$$\left\{ \underline{U} \right\} = \left\{ \underline{U} \right\}^+ - \left\{ \underline{U} \right\}^- \quad \dots (4.51)$$

with $\left\{ \underline{U} \right\}^+ \geq 0$ and $\left\{ \underline{U} \right\}^- \geq 0$

Therefore the above Eqn. (4.51) is expressed in the 'tableau' form as

$$\beta = \text{Min} \begin{array}{|c|c|c|c|} \hline \{\underline{U}\}^+ & \{\underline{U}\}^- & \{\underline{\Delta}\} & \\ \hline \{0\}^T & \{0\}^T & \{K_0\}^T & \\ \hline [G_0] & -[G_0] & -[\tilde{N}] & \{0\} \\ \hline \{0\}^T & \{0\}^T & \{\tilde{M}\} & 1 \\ \hline \end{array} \dots (4.52)$$

The dual problem, Eqn. (4.52), has as many variables as Eqn. (4.49), the primal problem, has number of constraints. The computation time is proportional only to the number of variables but vary to the cube of the number of constraints. Therefore, formulation as described by Eqn. (4.52) is more attractive from computational standpoint.

4.10 Dead Load

If the foundation is subjected to non-repeated loading such as gravity loading, T_D , below the bearing capacity of the foundation in addition to the variable repeated or cyclic loading, $\{\hat{T}\}$, the loading programme then may be written as

$$\{p_B(t)\} = \beta_s \{\hat{T}(t)\} + \{T_D\} \dots (4.53)$$

The energy balance of the external and internal work may be written as

$$\{K_0\}^T \{\underline{\Delta}\} = \beta_s \{\tilde{M}\}^T \{\underline{\Delta}\} + \int^T \{T_D\}^T \{\underline{\dot{U}}\} dt$$

or

$$\{K_o\}^T \{\underline{\Delta}\} - \{T_D\}^T \{\underline{U}\} = \beta_s \{\tilde{M}\}^T \{\underline{\Delta}\} \dots (4.54)$$

Therefore the modified 'tableau' of Eqn. (4.52) taking into account the effects of non cyclic loading (e.g submerged weight of the structure) becomes

$$\beta = \text{Min} \begin{array}{|c|c|c|c|} \hline \{\underline{U}\}^+ & \{\underline{U}\}^- & \{\underline{\Delta}\} & \\ \hline -\{T_D\}^T & \{T_D\}^T & \{K_o\}^T & \\ \hline [G_o] & -[G_o] & -[N] & \{0\} \\ \hline \{0\}^T & \{0\}^T & \{\tilde{M}\}^T & 1 \\ \hline \end{array} = \dots (4.55)$$

4.11 Summary of Steps to Evaluate the Shakedown Load Factor for the Offshore Foundation Problem

Step 1. Carry out the dynamic analysis with prescribed initial conditons, or static analysis depending on the nature of the problem for the prescribed load domain. For static response, consider the peaks of the loads (maxima), and for the dynamic response, adjust the initial conditions so that the response becomes periodic with respect to the cyclic wave loading.

Step 2. Maximize these stress response vectors with respect to the assumed linearized yield planes, so that the vector, $\{\tilde{M}\}$, may be constructed.

Step 3. Construct the global compatibility $[G_o]$ and gradient, $[\tilde{N}]$, matrices for the entire structure-foundation system. Also, compute the global $\{K_o\}$ vector based on the individual yield plane, ($i = 1, 2, \dots R$) and assembling over all elements ($j = 1, 2, \dots M$).

Step 4. Carry out the minimization problem based on the linear programming 'tableau' given by Eqn. (4.55).

4.12 Checking of Computer Programme

Two types of problems are chosen in order to compute the shakedown load factors based on the kinematic formulation given in Eqn. (4.55). These problems are i) a thick cylinder and ii) a strip footing. Both problems are analysed assuming a plane strain condition.

4.12.1 Shakedown Load For A Thick Cylinder

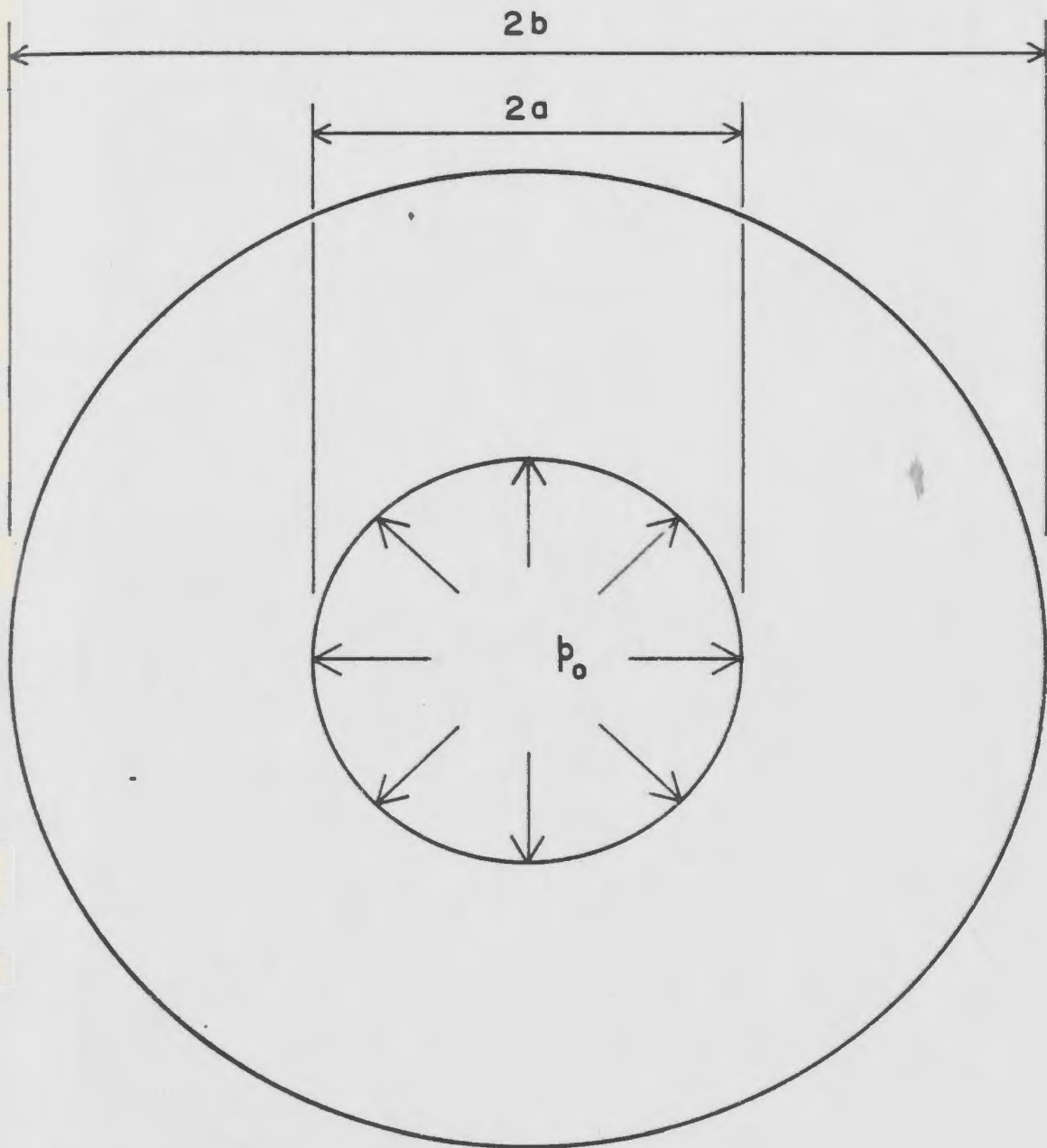
The thick cylinder shown in Fig. 4.5(a) is subjected to an internal pressure and has a ratio of outer to inner radius of 2.50. The quarter of the cylinder is discretized because of the symmetry. Three types of finite element meshes are considered for the analyses and shown in Figs. 4.5(b), 4.5(c) and 4.5(d). The internal pressure, P_o , is assumed to vary repeatedly between zero and some prescribed values, P^* .

The material properties used in all the analyses are $\frac{E}{\sigma_o} = 0.50 \times 10^3$, $\nu = 0.30$ and $\sigma_o = 0.418$ MPa where E = Young's modulus, ν = Poisson's ratio and σ_o = uniaxial yield stress in tension.

The material is assumed to be elastic-perfectly plastic and obey Von Mises yield condition. Six linearized planes are used to describe the yield surface.

The theoretical solutions for elastic and limit pressures are given by the following two equations (Hill, 1950) as

$$p_e = \frac{c}{\alpha_1} \left[3 \left(\frac{G}{G+3K_c} \right)^2 + \left(\frac{b}{a} \right)^4 \right]^{-1/2}$$



$$0 < p_o < p^*$$

FIG. 4.5(a) THICK CYLINDER UNDER REPEATED INTERNAL PRESSURE.

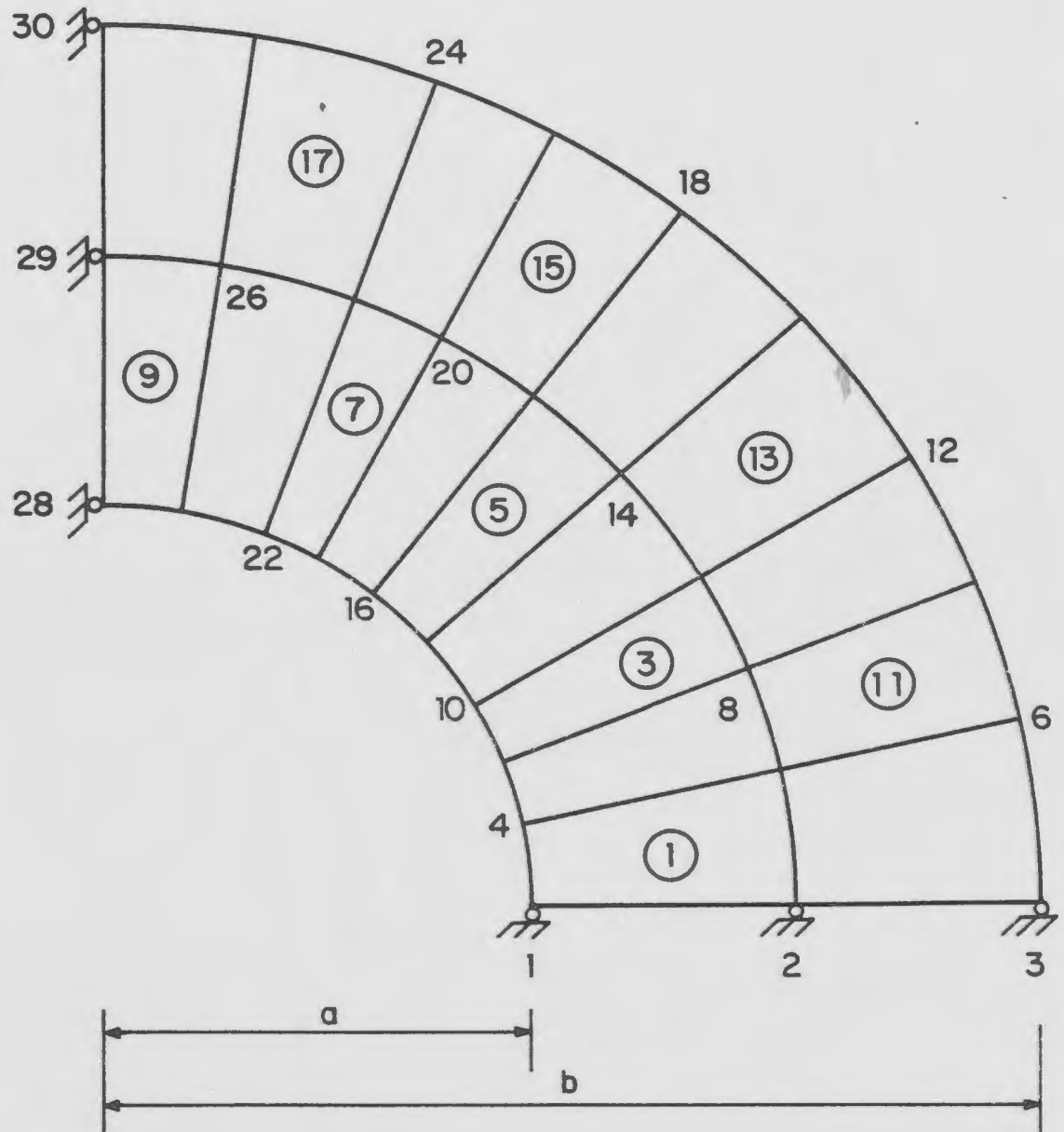


FIG. 4.5 (b) FINITE ELEMENT IDEALIZATION OF A THICK CYLINDER (18 ELEMENTS)

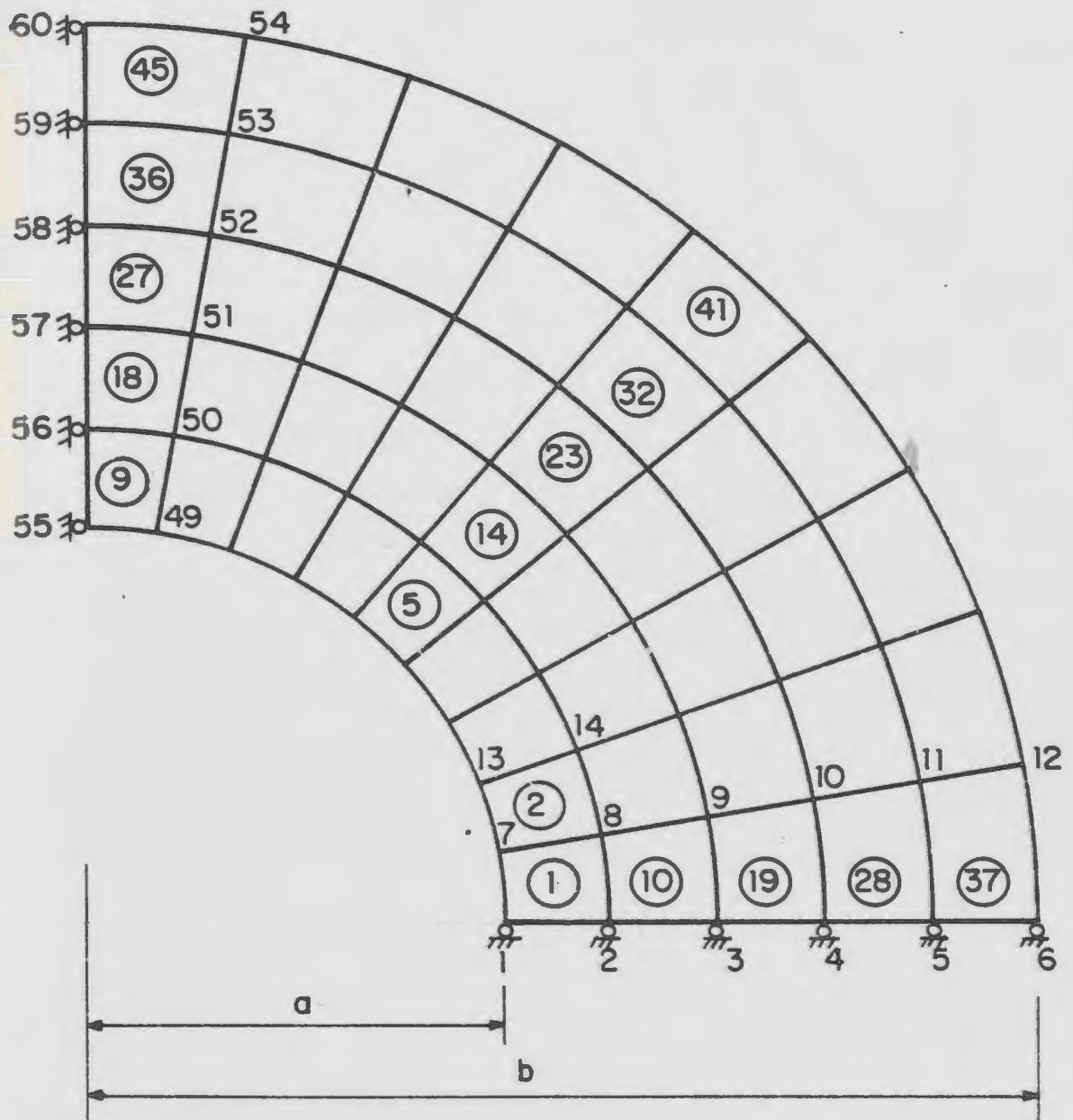


FIG. 4.5 (d) FINITE ELEMENT IDEALIZATION OF A THICK CYLINDER (45 ELEMENTS)

where

$$\alpha_1 = \frac{a^2}{(b^2 - a^2)}$$

and

$$\begin{aligned} p_c &= \text{limit pressure} \\ &= 2c \ln \left(\frac{b}{a} \right) \end{aligned}$$

where

b = outer radius

a = inner radius

$$c = \sigma_o / \sqrt{3}$$

σ_o = yield pressure for uniaxial tension

$$K_c = \text{bulk modulus} = \frac{E}{3(1-2\nu)}$$

$$G = \text{shear modulus} = \frac{E}{2(1+\nu)}$$

ν = Poisson's ratio

Because of the one parameter loading arrangement, theoretical limit pressure, p_c is equal to the shakedown pressure P_s . Table 4.1 provides limit pressures for three types of mesh arrangements. It is seen from Fig. 4.6 that as the number of elements increase the solution converges quite rapidly to the theoretical values. In Fig. 4.6 the elastic shakedown and collapse pressures are expressed in nondimensional form such as elastic pressure = P_e/c , shakedown pressure = P_s/c and collapse pressure = P_c/c .

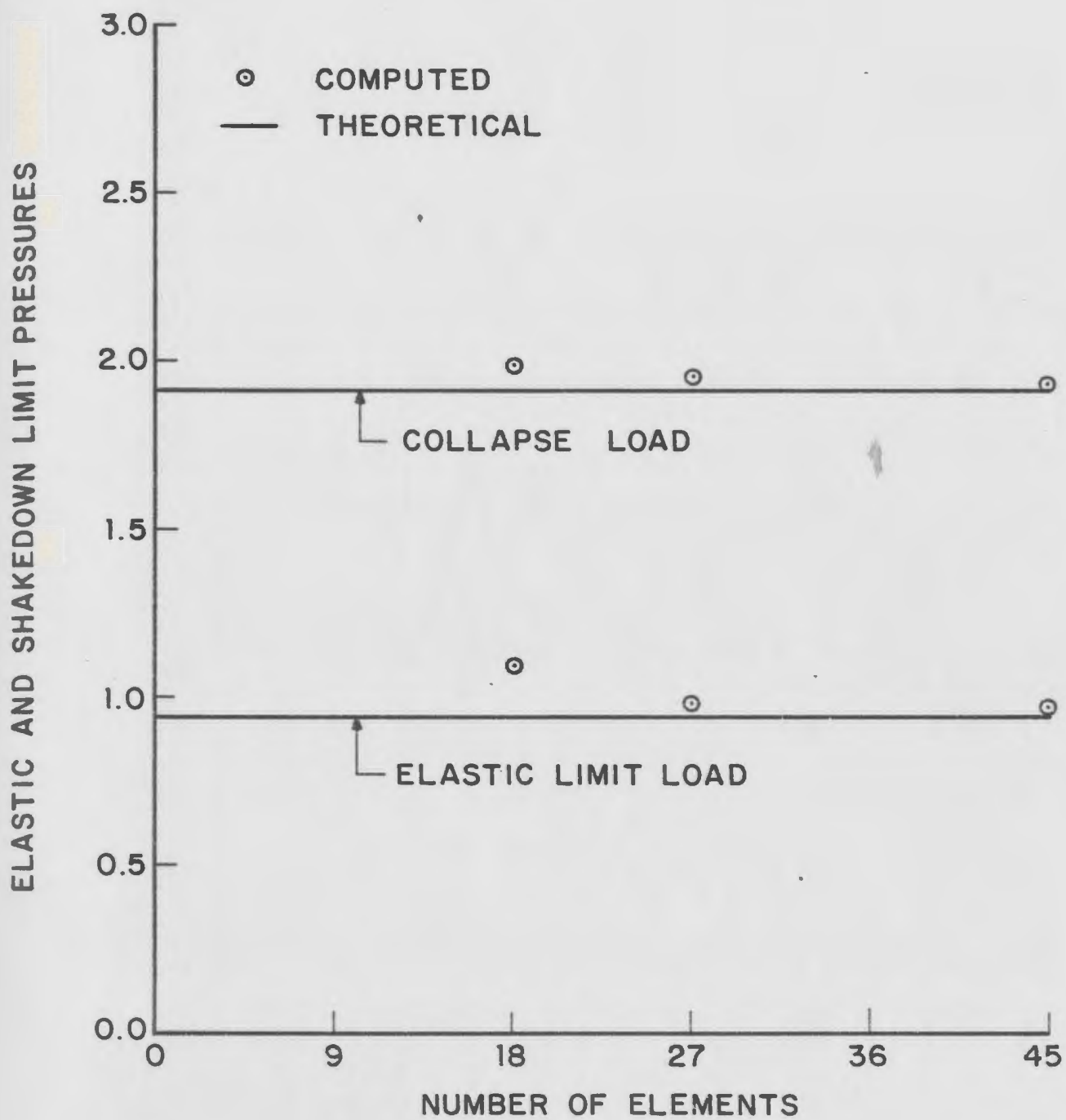


FIG. 4.6 INFLUENCE OF NUMBER OF ELEMENTS ON SHAKEDOWN LIMIT PRESSURE

TABLE 4.1 - INFLUENCE OF NUMBER OF ELEMENTS
ON SHAKEDOWN LIMIT PRESSURES

Number of Elements	Number of Nodal Points	Number of Equations	Elastic Pressure (p_e/c)	Shakedown Pressure (p_s/c)	Theoretical Collapse Pressure (p_c/c)
18	30	54	1.22	1.94	1.83
27	40	72	0.96	1.88	1.83
45	60	108	0.92	1.84	1.83

4.12.2 Shakedown Load For A Strip Footing

The application of the kinematic shakedown formulation as described before is illustrated by the case of a strip footing underlaid by a shallow stratum of clay under undrained condition. The loading magnitude is assumed to vary repeatedly downwards between zero and some prescribed value and is defined here by a dimensionless parameter, F , equal to the ratio of the applied load per unit area to the undrained shear strength (i.e. $F = \frac{W}{As_u}$ where W = load, A = area and s_u = undrained shear strength of the soil).

The soil medium is assumed to be homogeneous and isotropic. The finite element mesh of the strip footing is shown in Fig. 4.7. It consists of 45 nodes and 32 four noded rectangular isoparametric elements. The ratio of the footing width to the depth of the stratum is 0.80. Due to symmetry, only half of the footing geometry and the mesh is shown. Vertical side boundary is placed at a distance eight and a half times the width of the footing and assumed to be smooth. The bottom boundary is assumed to be rough and rigid. The footing is assumed to deform under plane strain condition. Reduced integration technique is used

to ensure the nearly incompressibility of the soil medium.

In the undrained analysis the soil is modelled as a weightless two-phase material with $s_u = 100$ KPa and $\phi_u = 0$, where s_u = undrained shear strength and ϕ_u = angle of internal friction. The drained elastic Lamé's parameters $\lambda = 150 s_u$, $\mu = 100 s_u$ and drained Poisson's ratio, $\nu_d = 0.30$ are assumed for the soil skeleton whereas the bulk modulus for pore water is assigned as $K_w = 2000$ MPa. This gives an equivalent undrained Poisson's ratio, $\nu_u \approx 0.499$ approximately ensuring the nearly incompressibility of the soil medium. The soil behaviour is assumed to be elastic-perfectly plastic and obey the Von Mises yield condition. Six linearized planes are used here to describe the yield condition.

The repeated vertical pressure is applied at nodes 1, 6 and 11 (refer Fig. 4.7). The stress fields comprising of σ_x , σ_y , σ_z and τ_{xy} for all the elements are obtained from which the elastic limit pressure is computed by observing the pressure at which any one of the elements becomes first plastic. The shakedown limit pressure is obtained by first normalizing the elastic stress responses for all

MESH GENERATION

NODAL POINT NUMBERING

ELEMENT NUMBERING

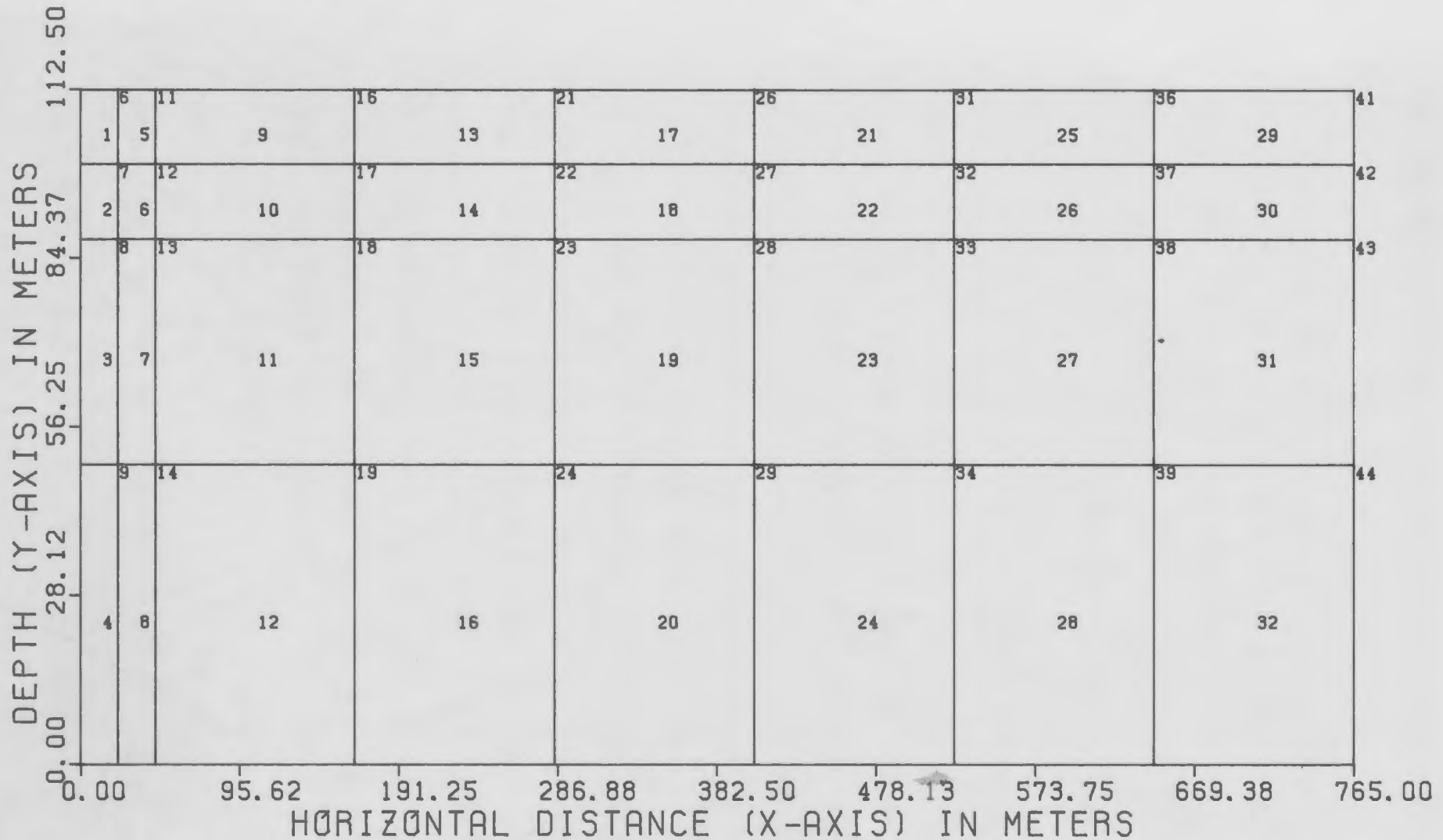


FIG. 4.7 FINITE ELEMENT MESH FOR STRIP FOUNDATION
(SHAKEDOWN ANALYSIS)

elements following Eqn. (4.43) and then solving the mathematical programming problem as given by Eqn. (4.55) using the maxima of these elastic responses. Total number of constraints and variables for this problem are 97 and 320 respectively. The typical CPU time required is 20 minutes and the major portion of this time is spent in solving the programming problem.

Table 4.2 shows the elastic and shakedown limit pressures in dimensionless form. Because of the one parameter loading situation, the shakedown pressure, F^S , will be equal to the collapse load, F^C , which is given theoretically as $F^C = (2+\pi)s_u$ in which F^C is the failure pressure (Hill, 1950). For a strip footing the above pressure is a true load because the applications of the two limit theorems (e.g. upper bound and lower bound) provide the same results. However, the shakedown pressure computed from Eqn. (4.55) will give an upper bound solution which is 4.65% higher than the theoretical collapse pressure. The elastic limit pressure, F^e , is only 4.50% higher than the theoretical value (Jumikis, 1969).

The strip footing problem is also run with a free lateral boundary and smooth rigid bottom. In this

case the first yield occurs at a lower applied pressure because of the reduction in horizontal normal stress. The computed value of the first yield in dimensionless form is 2.18 (Table 4.3). The shakedown pressure shown in Table 4.3 is 2.42. Hill (1950) computed the ultimate load for the same problem as 2.57. Therefore, the comparison shows excellent agreement with the published values thus ensuring the accuracy of the computer programme.

TABLE 4.2 - COMPUTATION OF ELASTIC AND SHAKEDOWN LIMIT PRESSURES FOR A STRIP FOOTING UNDER VARIABLE REPEATED LOADING

Footing Width Depth of Foundation	Number of Elements	Number of Nodal Points	Number of Equations	Elastic Limit Pressure (F^e/s_u)	Shakedown Limit Pressure (F^s/s_u)
0.80	32	45	64	3.28	5.38

TABLE 4.3 - COMPUTATION OF ELASTIC AND SHAKEDOWN PRESSURES FOR A STRIP FOOTING (HILL'S PROBLEM, 1950)

Footing Width Depth of Foundation	Number of Elements	Number of Nodal Points	Number of Equations	Elastic Limit Pressure (F^e/s_u)	Shakedown Limit Pressure (F^s/s_u)
0.80	32	45	77	2.18	2.42

CHAPTER V

DYNAMIC RESPONSE ANALYSIS
USING NONLINEAR SOIL MODELS5.1 Introduction

This chapter presents the formulation of three different types of soil constitutive relationships which will be used in the dynamic response analysis of a gravity type offshore foundation. Research in the field of earthquake engineering has shown that the soil deformation can vary from small to very large amplitude depending on the nature of the dynamic loading. Cyclic loading causes permanent deformation and thus strain, under both drained and undrained conditions. The permanent strains are cumulative in nature i.e. carried over from one cycle to another and remain even if the cyclic loading has been terminated. The amount of straining depends primarily on the soil type, average effective confining pressure, void ratio, degree of saturation, frequency of vibration, over consolidation ratio (OCR), etc;

The chapter is divided in three sections. The first section describes the formulation of the three types

of soil modellings: These are a) elastic b) equivalent linear and c) incremental nonlinear elastic-plastic. Although the equivalent linear model is quite popular in the field of earthquake engineering, it fails to provide any information on the permanent deformation which is of great concern to the designer of an offshore platform. Nevertheless, the analysis can still provide some valuable information on strain compatible soil properties which will provide a realistic soil stress distribution. On the other hand, the incremental elastic-plastic analysis coupled with direct time step integration approach is capable of predicting the permanent deformation history. However, in an incremental stress-strain relationship, the solution has to proceed in small time steps thus limiting the analysis to a very few number of cycles.

The second section presents the incremental formulation of dynamic equations of motion in time domain. A computer code has been developed for the dynamic response analysis of an offshore gravity foundation based on the above soil modellings. An example problem is illustrated with particular reference to the prediction of permanent deformation of a strip footing under

repeated vertical loading condition.

5.2 Soil Models

The mechanical behaviour of saturated soils leads to very complicated stress-strain relationship. Therefore, the successful modelling must at least consider the two-phase characteristics where the behaviour is primarily affected by the hydrostatic pressure, fully controlled by the effective stresses, the applied total stress and its past history. The undrained behaviour of the soil-foundation system is given by Eqn. (3.38j) where, $[K_{ss}]$, represents the undrained stiffness matrix and expressed as

$$K_{ss} = \int_V \psi_{11}^T D_{sd} \psi_{11} dV + \int_V \alpha^2 M_o \psi_{22}^T \psi_{22} dV$$

If the stress-strain relationship cannot be expressed uniquely and is somewhat dependent on loading history or strain, the constitutive matrix, $[D_{sd}]$, must be updated in order to make it either strain compatible for equivalent linear model or function of existing state of stress incremental nonlinear model. However, for an elastic model, $[D_{sd}]$, remains constant during the analysis.

5.2.1 Linear Elastic Model

Among the soil models proposed in the literature, the elastic model is the simplest, where the stress-strain relations are reversible and have no time-delay. The elastic model still can provide some useful results, which gives the starting point of developing other more complex models. The elastic response analysis requires the elastic properties of the soil and structure e.g. usually defined by Lamé's constants, as the input.

5.2.2 Equivalent Linear Model

The equivalent linear model is used here to study the dynamic response of the soil-foundation system to wave excitation. The method was originally developed for the dynamic response analysis of soil deposits to seismic excitation. The basic assumptions which were made in formulating the equivalent linear model were: a) seismic excitation is primarily due to shear waves propagating vertically and b) the strong nonlinear characteristics of the soil may be modelled by a viscously damped linear oscillator whose properties such as stiffness and damping are strain dependent.

Extensive experimental research on dynamic soil properties has shown that dynamic soil properties such as shear modulus and damping are strain dependent. The shear modulus decreases with increasing strain amplitude whereas damping increases with the strain amplitude of the loading as shown in Fig. 5.1.

From the extensive series of reversed torsional tests on cohesive and cohesionless soils and the information already available in the literature, it is reasonable to assume that a modified hyperbolic curve will satisfactorily represent the shearing stress-shearing strain relationship throughout the range of strain amplitude up to failure. Fig. 5.2 shows the factor which governs the basic hyperbolic shearing stress-strain curve. At zero shearing strain, the tangent to the curve represents the maximum shear modulus, G_{\max} . The secant modulus at any point along the curve for example, point A is designed as, G , and the maximum shearing stress as, τ_{\max} , (determined from a simple shear test). The horizontal line at the ordinate of, τ_{\max} , is the second asymptote to the hyperbolic curve as shown in Fig. 5.2. G_{\max} and τ_{\max} can be obtained from field and/or laboratory tests.

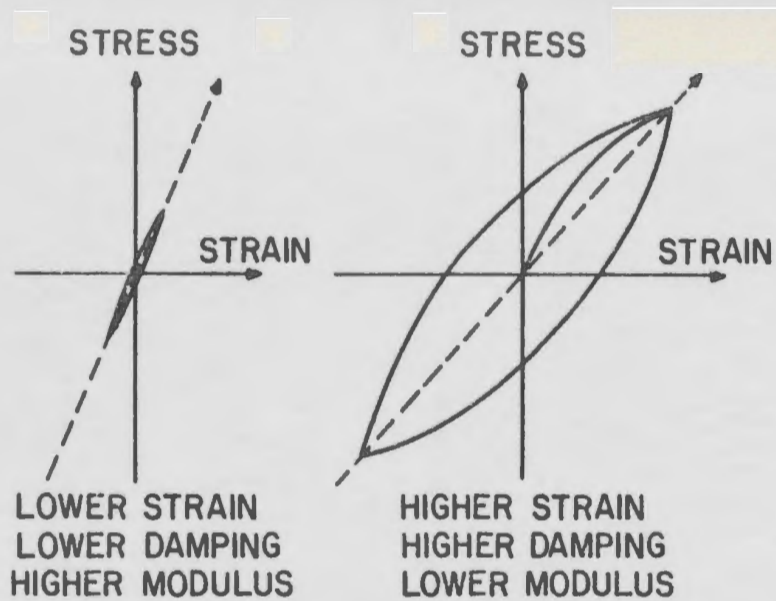


FIG. 5.1 ILLUSTRATION OF STRAIN DEPENDENCY
OF MODULI AND DAMPING IN SOILS
(SEED AND IDRIS, 1969)

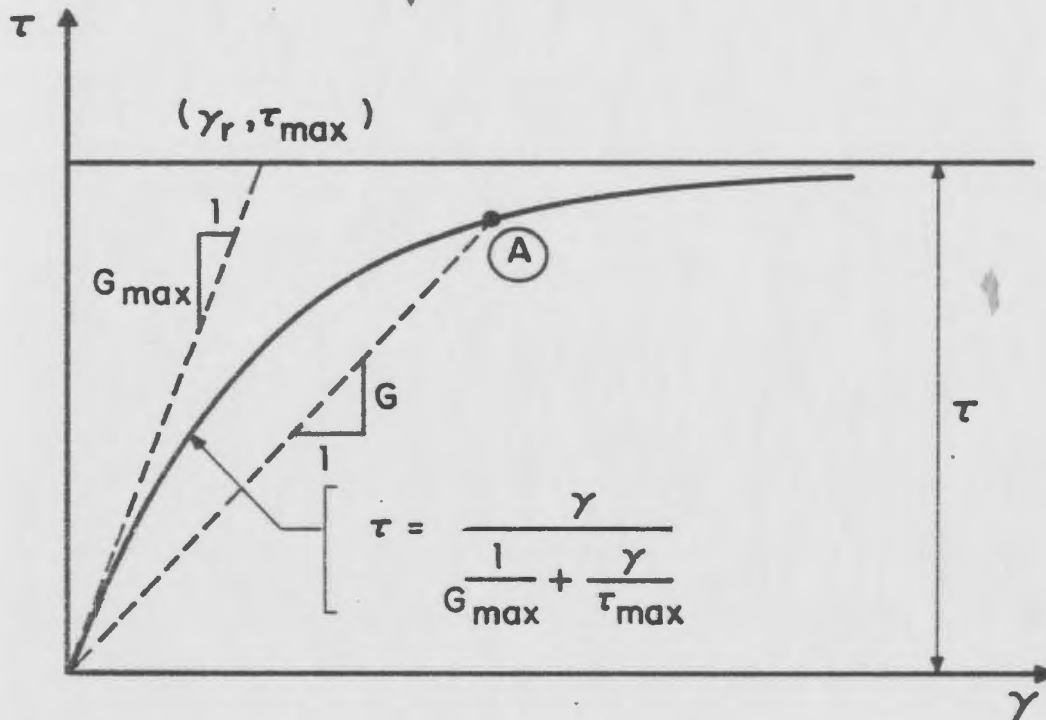


FIG. 5.2 BASIC PARAMETERS FOR HYPERBOLIC SHEARING STRESS-SHEARING STRAIN CURVES (SEED AND IDRIS, 1969)

5.2.2.1 Variable Stiffness and Damping Solution

The basic procedure adopted in an equivalent linear model is that the dynamic response of the soil-foundation system may be approximated sufficiently by a damped linear viscous oscillator if the properties of this oscillator (e.g. stiffness and damping) are chosen properly. An iterative method is normally used in order to derive these properties. While using this method in the finite element analysis, the stress-strain characteristics of the soil are assigned in each element with the following four properties; a) maximum shear modulus, G_{\max} , defined as the shear modulus at low strain $10^{-4}\%$ level b) maximum damping, D_{\max} , (= 29%) c) Poisson's ratio and d) estimated shear modulus, G , for the first iteration. The average in-situ shear moduli for saturated clays are shown in Fig. 5.3. These curves are also tabulated in Table 5.1 and are shown in Fig. 5.4 in terms of G/G_{\max} and damping ratios, D .

In the first cycle of the analysis, initial values of shear moduli and damping ratios are chosen and a dynamic linear elastic response analysis is performed for the entire time

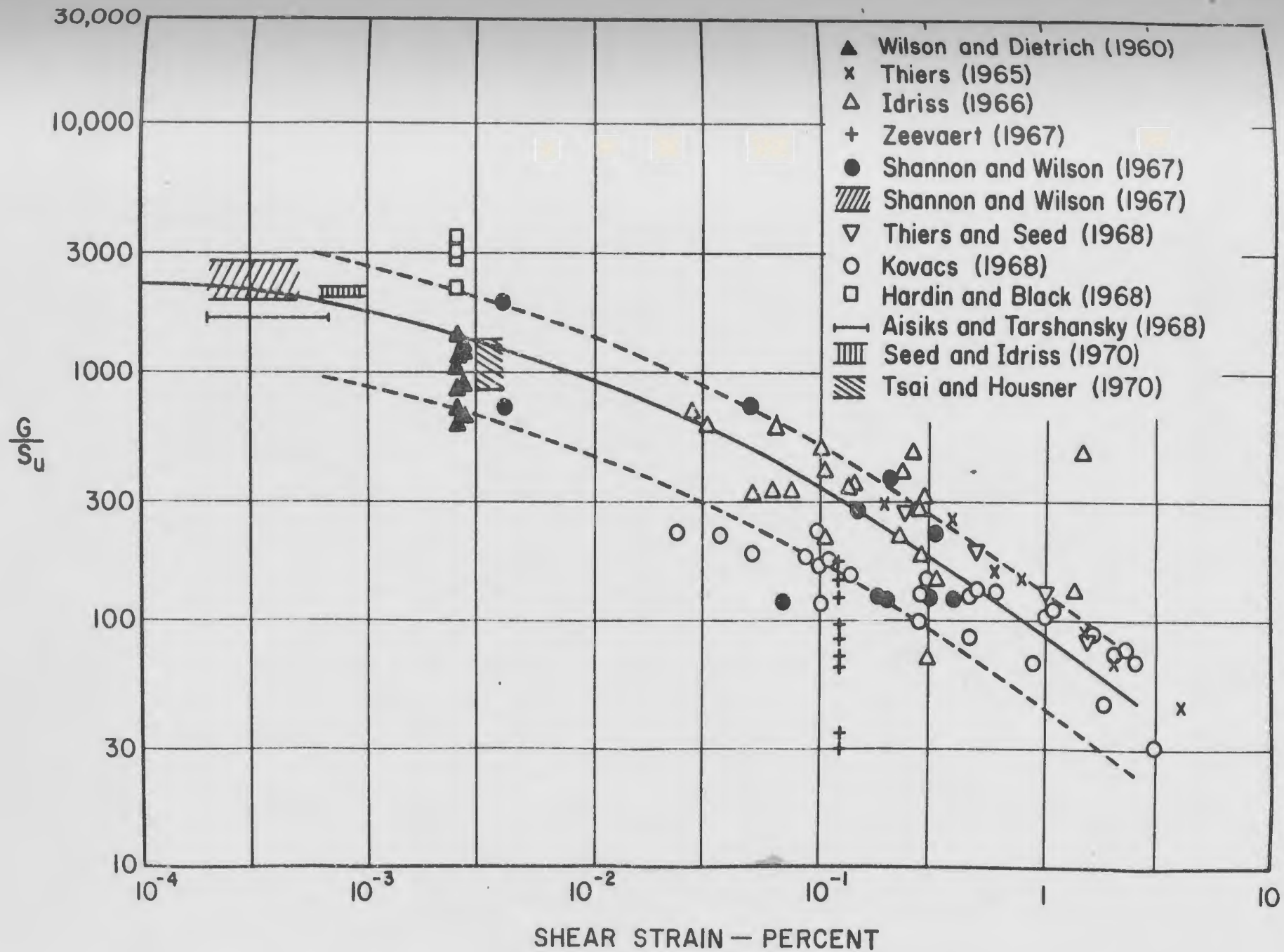


FIG. 5.3 IN-SITU SHEAR MODULI FOR SATURATED CLAYS
(SEED AND IDRIS, 1969)

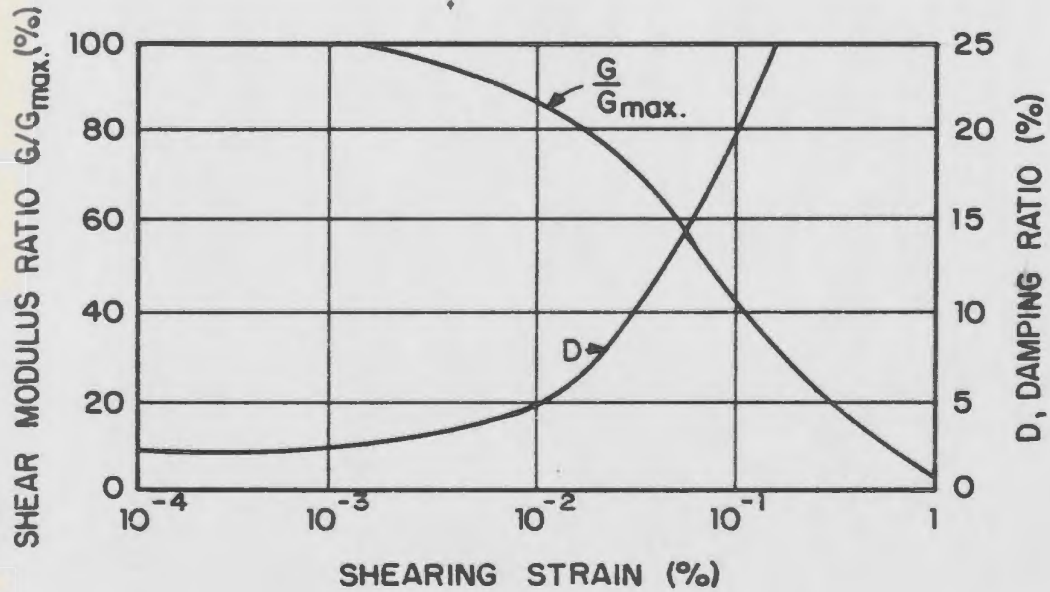


FIG. 5.4 MODULUS AND DAMPING RATIOS USED FOR DYNAMIC RESPONSE ANALYSIS
(SEED AND IDRIS, 1969)

TABLE 5.1 - STRAIN-COMPATIBLE SOIL PROPERTIES
(SEED AND IDRIS, 1969)

Effective Shear Strain (%)	$\log (\gamma_{\text{eff}})$	Shear Modulus Reduction Factor*	Fraction of Critical Damping (%)
		Clay	Clay
1.0×10^{-4}	-4.0	1.000	2.50
3.16×10^{-4}	-3.5	0.913	2.50
1.00×10^{-3}	-3.0	0.761	2.50
3.16×10^{-3}	-2.5	0.565	3.50
1.00×10^{-2}	-2.0	0.400	4.75
3.16×10^{-2}	-1.5	0.261	6.50
1.00×10^{-1}	-1.0	0.152	9.25
0.316	-0.5	0.076	13.8
1.00	0.	0.037	20.0
3.16	0.5	0.013	26.0
10.00	1.0	0.004	29.0

* This is the factor which has to be applied to the shear modulus at low shear strain amplitudes (here defined as 10^{-4} percent) to obtain the modulus at higher strain levels.

histories of the wave force which may contain several number of cycles. The maximum shear strain, γ_{\max} , is selected for each element in order to compute the new strain compatible shear modulus and damping ratio. Having computed the new value of shear modulus and damping ratio for each element depending on the maximum shear strain level, a second analysis is carried out for the next iteration. The procedure is repeated until no significant changes in moduli or damping ratios are observed for all elements. The final response obtained from the last iteration is assumed to be a reasonable approximation for the true nonlinear response.

In the variable stiffness and damping solution, the stiffness and damping submatrices are formed for each element and then all element submatrices are assembled in global form in a conventional manner. The global stiffness and mass matrices are computed and an eigenvalue analysis is performed. In order to compute the damping submatrix of an element, j , the Rayleigh type damping is assumed. For an element, j ,

$$[c]_j = B_1[m]_j + B_2[\bar{k}]_j \quad \dots (5.1)$$

in which $[c]_j$, $[m]_j$ and $[\bar{k}]_j$ are the damping, mass and stiffness submatrices respectively and B_1 and B_2 are parameters which are functions of the damping ratios and stiffness characteristics. The Rayleigh parameters B_1 and B_2 are expressed as

$$B_1 = D_j \omega_1 \quad \dots (5.2)$$

and $B_2 = D_j / \omega_1$

where, D_j , represents the strain dependent damping ratio for element j , obtained from Fig. 5.4 and Table 5.1 and, ω_1 , represents the fundamental frequency of the soil-foundation system. $[m]$ and $[\bar{k}]$ are given by Eqn. (3.36).

The global damping matrix, C^* , for the entire assemblage is formed by appropriate addition of the damping submatrices of all finite elements in the discretized model. Thus if, c_{k1}^j , represents $(k1)$ -th term of the damping submatrix $[c]_j$ of a typical element j , the $k1$ -th term of the damping matrix of the entire system of M -elements is given by

$$C_{k1}^* = \sum_{j=1}^M c_{k1}^j \quad \dots (5.3)$$

The resulting damping matrix is symmetric and normally sparsely populated. The response (displacement) from Eqn. (3.48) is obtained by a step-by-step integration procedure described in the next section. Haldar et al (1980) have used the equivalent linear model in the dynamic response analysis of a submarine pipe line to seismic excitation and observed the significant effect on the stresses in the porous bed around the pipe.

5.2.3 Elastic Plastic Model

The primary assumption of the incremental elastic-plastic soil model is that the total strain increments are separable into elastic and plastic components, in which the elastic part is linearly related to the stress increment. The plastic component of the strain increment is defined by means of a plastic potential, g , and yield functions, f_y . The stress level at which plastic strain develops is defined by the yield function whereas the plastic potential function defines the distribution of the plastic strain increment during the plastic flow. The magnitude of the plastic strain is determined by a particular work-hardening relationship. If the

yield surface is defined by a continuous differentiable function, f_y , the three possible loading states, which will be encountered during an incremental load path, are as follows:

- 1) During the loading state the stress increments are directed outwards from the yield surface and as such

$$\frac{\partial f_y}{\partial \sigma_{ij}} d\sigma_{ij} > 0 \quad \dots (5.4a)$$

- 2) During the neutral loading state, the stress increments will remain tangent to the yield surface and as such

$$\frac{\partial f_y}{\partial \sigma_{ij}} d\sigma_{ij} = 0 \quad \dots (5.4b)$$

and

- 3) During the unloading state, the stress increments are directed inwards and as such

$$\frac{\partial f_y}{\partial \sigma_{ij}} d\sigma_{ij} \leq 0 \quad \dots (5.4c)$$

In the associative flow theory of plasticity, the yield function, f_y , is assumed to coincide with the plastic potential function, g . In deriving the elastic-plastic stress-strain relationship in an incremental form, a single continuous yield

function, f_y , is assumed and the total strain, $d\epsilon_{ij}$, given by

$$d\epsilon_{ij} = d\epsilon_{ij}^E + d\epsilon_{ij}^P \quad \dots (5.5)$$

where

$d\epsilon_{ij}^E$ = elastic strain component

$d\epsilon_{ij}^P$ = plastic strain component

and $d\epsilon_{ij}$ = total strain

The strain increments for three dimensional stress state is represented in vector form as

$$\{d\epsilon_{ij}\}^T = \langle d\epsilon_x \quad d\epsilon_y \quad d\epsilon_z \quad d\gamma_{xy} \quad d\gamma_{yz} \quad d\gamma_{zx} \rangle \quad \dots (5.6a)$$

and in two dimensional stress state

$$\{d\epsilon_{ij}\}^T = \langle d\epsilon_x \quad d\epsilon_y \quad d\epsilon_z \quad d\gamma_{xy} \rangle \quad \dots (5.6b)$$

5.2.3.1 Elastic Strain Increment

The elastic strain increments are related to the effective stress increments as

$$\{d\sigma_{ij}\} = [D_{sd}] \{d\epsilon_{ij}^E\} \quad \dots (5.7)$$

where $\{d\sigma'_{ij}\}$ = effective stress increment
vector

and

$[D_{sd}]$ = drained elastic stress-strain
matrix as defined by Eqn.
(3.38i)

5.2.3.2 Plastic Strain Increment

The initial and subsequent yield condition for any work hardening model can be written as

$$f_y (\sigma'_{ij}, \epsilon_{ij}^p, b) = 0 \quad \dots (5.8)$$

in which, σ'_{ij} , ϵ_{ij}^p , and b are the effective stress components, the plastic strain components and a material constant respectively. The constant, b , is a function of effective plastic strain, $\bar{\epsilon}^p$.

The flow rule, which relates the relative magnitude of the plastic strain increments to the stresses, is

$$d\epsilon_{ij}^p = \Delta \frac{\partial f_y}{\partial \sigma'_{ij}} \quad \dots (5.9)$$

and the equivalent plastic strain

$$d\bar{\epsilon}^P = \Delta \sqrt{d\epsilon_{ij}^P \cdot d\epsilon_{ij}^P} \quad \dots (5.10)$$

The vector of stress gradient is expressed as

$$\left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\}^T = \left[\begin{array}{cccc} \frac{\partial f_y}{\partial \sigma'_x} & \frac{\partial f_y}{\partial \sigma'_y} & \frac{\partial f_y}{\partial \sigma'_z} & \frac{\partial f_y}{\partial \tau_{xy}} \end{array} \right] \quad \dots (5.11)$$

5.2.3.3 Work Hardening Law

The work hardening law determines the actual magnitude of the plastic strain increment caused by a given stress increment. Therefore, the work hardening law is defined here as the relationship between the effective stress level and the plastic work done per unit volume and expressed as a function of equivalent plastic work as, $Q(\epsilon_{ij}^P)$ and in matrix form.

$$Q(\epsilon_{ij}^P) = \left\{ \sigma'_{ij} \right\}^T \left\{ \epsilon_{ij}^P \right\} \quad \dots (5.12)$$

5.2.3.4 Incremental Formulation

The incremental method assumes that during yielding, the function, f_y , should satisfy the consistency condition and therefore

$$df_y = 0$$

or

$$\left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\}^T \left\{ d\sigma'_{ij} \right\} + \left\{ \frac{\partial f_y}{\partial \epsilon^p_{ij}} \right\}^T \left\{ d\epsilon^p_{ij} \right\} + \frac{\partial f_y}{\partial \bar{e}^p} d\bar{e}^p = 0 \quad \dots (5.13)$$

where

$$\left\{ \frac{\partial f_y}{\partial \epsilon^p_{ij}} \right\}^T = \left[\begin{array}{cccc} \frac{\partial f_y}{\partial \epsilon^p_x} & \frac{\partial f_y}{\partial \epsilon^p_y} & \frac{\partial f_y}{\partial \epsilon^p_z} & \frac{\partial f_y}{\partial \gamma^p_{xy}} \end{array} \right] \quad \dots (5.14)$$

Using the elastic strain to effective stress relationship as defined by Eqn. (5.7), Eqn. (5.13) becomes

$$\left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\}^T [D_{sd}] \left\{ d\epsilon^E_{ij} \right\} + \left\{ \frac{\partial f_y}{\partial \epsilon^p_{ij}} \right\}^T \left\{ d\epsilon^p_{ij} \right\} + Y = 0 \quad \dots (5.15)$$

$$\text{where } Y = \frac{\partial f_y}{\partial \bar{e}^p} d\bar{e}^p$$

Substituting Eqn. (5.5) in Eqn. (5.15), one obtains

$$\left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\}^T [D_{sd}] \left(\left\{ d\epsilon_{ij} \right\} - \left\{ d\epsilon^p_{ij} \right\} \right) + \left\{ \frac{\partial f_y}{\partial \epsilon^p_{ij}} \right\}^T \left\{ d\epsilon^p_{ij} \right\} + Y = 0 \quad \dots (5.16)$$

Substituting the flow rule from Eqn. (5.9) into Eqn. (5.16), one obtains

$$\left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\}^T [D_{sd}] (\{d\epsilon_{ij}\} - \Delta \left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\}) + \left\{ \frac{\partial f_y}{\partial \epsilon^p_{ij}} \right\}^T \left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\} + Y = 0 \quad \dots (5.17)$$

Writing the Eqn. (5.17) in the following form, we get

$$\left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\}^T [D_{sd}] (\{d\epsilon_{ij}\} - \Delta \left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\}) + H\Delta = 0 \quad \dots (5.18)$$

where

$$H = \left\{ \frac{\partial f_y}{\partial \epsilon^p_{ij}} \right\}^T \left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\} + \frac{\partial f_y}{\partial \epsilon^p} \sqrt{d\epsilon^p_{ij} \cdot d\epsilon^p_{ij}} \quad \dots (5.19)$$

Solving for Δ from Eqn. (5.18), one obtains

$$\Delta = \frac{\left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\}^T [D_{sd}] \{d\epsilon_{ij}\}}{\left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\}^T [D_{sd}] \left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\} - H} \quad \dots (5.20)$$

Substituting the value of Δ in Eqn. (5.7), one obtains

$$\begin{aligned} \{d\sigma'_{ij}\} &= [D_{sd}] \left(\{d\epsilon_{ij}\} - \Delta \left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\} \right) \\ &= \left([D_{sd}] - \frac{[D_{sd}] \left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\} \left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\}^T [D_{sd}]}{\left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\}^T [D_{sd}] \left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\} - H} \right) \{d\epsilon_{ij}\} \end{aligned} \quad \dots (5.21)$$

The term in the bracket (.) is defined as the desired elastic plastic stress-strain matrix, which is symmetric because of the associative nature of flow rule. If, H , is zero, no work hardening is present and therefore the assumption of elastic-perfectly plastic model is valid. The strain increment is defined as

$$\{d\bar{\epsilon}\} = \bar{\Delta} \left\{ \frac{\partial f_y}{\partial \sigma'} \right\} \quad \dots (5.22)$$

where $\bar{\Delta}$ is arbitrary

and

$$\begin{aligned} \{d\sigma\} &= [D_{sd}] \{d\bar{\epsilon}\} - [D_{sd}] \left\{ \frac{\partial f_y}{\partial \sigma'} \right\} [L]^{-1} \left\{ \frac{\partial f_y}{\partial \sigma'} \right\}^T \\ &\quad [D_{sd}] \left\{ \frac{\partial f_y}{\partial \sigma'} \right\} \bar{\Delta} \end{aligned} \quad \dots (5.23a)$$

where

$$[L] = \begin{Bmatrix} \frac{\partial f_y}{\partial \sigma'} \\ \frac{\partial f_y}{\partial \sigma'} \end{Bmatrix}^T [D_{sd}] \begin{Bmatrix} \frac{\partial f_y}{\partial \sigma'} \\ \frac{\partial f_y}{\partial \sigma'} \end{Bmatrix} - H \quad \dots (5.23b)$$

If $H = 0$, one obtains

$$\{d\sigma\} = [D_{sd}]\{d\bar{\epsilon}\} - [D_{sd}] [L]^{-1} [L] \{d\bar{\epsilon}\} = 0 \quad \dots (5.24)$$

which implies that the plastic deformation will progress at constant stress level.

5.2.3.5 Drucker Prager Model

The Drucker Prager yield criteria is quite often used in the elastic-plastic finite element analysis and may be considered as a first attempt to approximate the well-known Mohr Coulomb failure criteria by a smooth surface (Fig. 5.5). The yield criteria is expressed in terms of the two stress invariants (first and second) together with two material constants a and b . The yield function is expressed as

$$f_y = a J_1 + \sqrt{J_2} - b = 0 \quad \dots (5.25a)$$

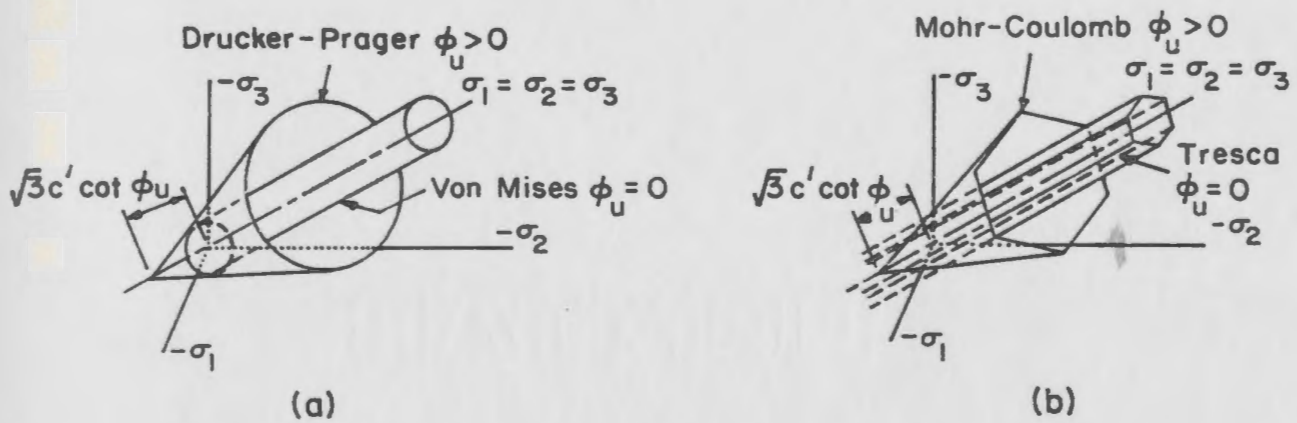


FIG. 5.5 TYPICAL YIELD FUNCTIONS IN GENERALISED STRESS-SPACE (ZIENKIEWICZ, 1977)

The constants a and b are called the material parameters which are related, in the case of plane strain, to the Coulomb's constants, c' and ϕ_u as

$$a = \frac{\sin^2 \phi_u}{\sqrt{3} (3 + \sin^2 \phi_u)} \quad \dots (5.25b)$$

and

$$b = \frac{(\sqrt{3}c' \cos \phi_u)^2}{3 + \sin^2 \phi_u} \quad \dots (5.25c)$$

The characteristics of undrained clays satisfying 'zero volume change' condition can be modelled mathematically by Von Mises yield criteria. This model is only applicable to undrained clays which have a constant shear strength. The undrained shear strength of clay, s_u , is then simply related to second invariant of stress, J_2 .

The equation of the Von Mises yield surface is expressed as

$$f_y = \sqrt{J_2} - b = 0 \quad \dots (5.25d)$$

The terms J_1 and J_2 are the first and second invariant of stresses and expressed as

$$J_1 = \sigma'_{ij} = (\sigma'_x + \sigma'_y + \sigma'_z) \quad \dots (5.26a)$$

$$\begin{aligned}
 J_2 &= 1/2 S_{ij} S_{ij} = 1/2 (S_x^2 + S_y^2 + S_z^2) + \tau_{xy}^2 \\
 &= 1/6 (\sigma'_x - \sigma'_y)^2 + (\sigma'_y - \sigma'_z)^2 + (\sigma'_z - \sigma'_x)^2 + 6\tau_{xy}^2 \\
 &\dots (5.26b)
 \end{aligned}$$

$$\text{where } S_x = \sigma'_x - \frac{J_1}{3}$$

$$S_y = \sigma'_y - \frac{J_1}{3} \dots (5.26c)$$

$$\text{and } S_z = \sigma'_z - \frac{J_1}{3}$$

The stress tensor, σ'_{ij} , can then be expressed in terms of J_1 and S_{ij} as

$$\sigma'_{ij} = 1/3 J_1 \delta_{ij} + S_{ij} = \sigma'_{mm} + S_{ij} \dots (5.26d)$$

where, δ_{ij} , is Kronecker delta. The mean hydrostatic pressure, σ'_{mm} , causes volumetric change whereas the deviatoric stress, S_{ij} , causes shape change in a soil element.

For perfect plasticity, b , is constant and can be expressed uniquely. However, for isotropic hardening, b , is not constant and is a function of accumulated increments of effective plastic strain

and expressed as

$$b = b (\bar{e}^p) \quad \dots (5.27)$$

where \bar{e}^p = effective plastic strain

For yielding under an uniaxial state of stress on a soil sample in vertical direction, σ_y , is active and all other stress components are zero. The yield function given by Eqn. (5.25a) becomes

$$a\sigma'_y + \frac{\sigma'_y}{\sqrt{3}} - b = 0$$

or

$$b = \left(a + \frac{1}{\sqrt{3}} \right) \sigma'_y \quad \dots (5.28)$$

Substituting Eqn. (5.28) in Eqn. (5.25a), the effective stress, $\bar{\sigma}$, is expressed as

$$\bar{\sigma} = \frac{aJ_1 + \sqrt{J_2}}{a + \frac{1}{\sqrt{3}}} \quad \dots (5.29)$$

The yielding occurs when $\bar{\sigma} = \sigma'_y$. By putting $a = 0$, one obtains the Von Mises yield condition and the effective stress.

The incremental effective plastic work is given by

$$\begin{aligned}
 dQ(\epsilon_{ij}^p) &= \left\{ \sigma'_{ij} \right\}^T d\epsilon_{ij}^p \\
 &= \bar{\sigma} d\bar{\epsilon}^p \quad \dots (5.30)
 \end{aligned}$$

Using Eqn. (5.25) and Eqn. (5.26), it can be shown that

$$\begin{aligned}
 \left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\} &= \frac{\partial f_y}{\partial J_1} \cdot \left\{ \frac{\partial J_1}{\partial \sigma'_{ij}} \right\} + \frac{\partial f_y}{\partial J_2} \cdot \left\{ \frac{\partial J_2}{\partial \sigma'_{ij}} \right\} \\
 &= a\delta_{ij} + S_{ij}/2 \sqrt{J_2} \quad \dots (5.31)
 \end{aligned}$$

Substituting Eqn. (5.9), (5.29) and (5.31) in Eqn. (5.30) and using the relationship

$$\sigma'_{ij} S_{ij} = \left(S_{ij} + \frac{\sigma_{mm}'}{3} \delta_{ij} \right) S_{ij} = 2J_2 \quad \dots (5.32)$$

one obtains

$$\begin{aligned}
 dQ(\epsilon_{ij}^p) &= \bar{\sigma} d\bar{\epsilon}^p = \Delta \left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\}^T \sigma'_{ij} \\
 &= \Delta \left(a\delta_{ij} + \frac{S_{ij}}{2\sqrt{J_2}} \right) \sigma'_{ij} \\
 &= \Delta \left(aJ_1 + \sqrt{J_2} \right) \\
 &= \Delta \left(a + \frac{1}{\sqrt{3}} \bar{\sigma} \right) \quad \dots (5.33)
 \end{aligned}$$

From Eqn. (5.9) one may write

$$\begin{aligned} \left\{d\epsilon_{ij}^p\right\} \cdot \left\{d\epsilon_{ij}^p\right\} &= \Delta^2 \left\{\frac{\partial f_y}{\partial \sigma'_{ij}}\right\} \cdot \left\{\frac{\partial f_y}{\partial \sigma'_{ij}}\right\} \\ &= \Delta^2 (3a^2 + 1/2) \quad \dots (5.34a) \end{aligned}$$

Therefore

$$\sqrt{\left\{d\epsilon_{ij}^p\right\} \cdot \left\{d\epsilon_{ij}^p\right\}} = \Delta \sqrt{(3a^2 + 1/2)} \quad \dots (5.34b)$$

From Eqn. (5.33)

$$\begin{aligned} d\bar{e}^p &= \Delta \left(a + \frac{1}{\sqrt{3}}\right) \text{ and using Eqn. (5.34b) one obtains} \\ d\bar{e}^p &= \sqrt{\left\{d\epsilon_{ij}^p\right\} \cdot \left\{d\epsilon_{ij}^p\right\}} \frac{\left(a + \frac{1}{\sqrt{3}}\right)}{\sqrt{3a^2 + 1/2}} \quad \dots (5.34c) \end{aligned}$$

Finally if $a = 0$, one obtains the Von Mises effective plastic strain increment from Eqn. (5.10).

The elastic plastic constitutive relationship expressed by Eqn. (5.21) can be presented after using Eqn. (5.31) and Eqn. (5.32) as follows

$$\begin{aligned} \left\{d\sigma'_{ij}\right\} &= \left[D_{sd}\right] \left(1 - \frac{\left[a\delta_{ij} + \frac{S_{ij}}{2J_2} \right] \left[a\delta_{ij} + \frac{S_{ij}}{2J_2} \right]^T \left[D_{sd}\right]}{\left[a\delta_{ij} + \frac{S_{ij}}{2J_2} \right]^T \left[D_{sd}\right] \left[a\delta_{ij} + \frac{S_{ij}}{2J_2} \right] - H} \right) \left\{d\epsilon_{ij}\right\} \\ &= \left[D_{sd}^{Ep}\right] \left\{d\epsilon_{ij}\right\} \quad \dots (5.35) \end{aligned}$$

where $[D_{sd}^{Ep}]$ is the desired elastic-plastic constitutive matrix based on effective stress vector.

From Eqn. (5.19)

$$\begin{aligned} H &= \left\{ \frac{\partial f_y}{\partial \epsilon_{ij}^p} \right\}^T \left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\} + \frac{\partial f_y}{\partial \bar{e}^p} \sqrt{d\epsilon_{ij}^p \cdot d\epsilon_{ij}^p} \\ &= \left\{ \frac{\partial f_y}{\partial \epsilon_{ij}^p} \right\}^T \left\{ \frac{\partial f_y}{\partial \sigma'_{ij}} \right\} + \frac{\partial f_y}{\partial b} \cdot \frac{\partial b}{\partial \bar{e}^p} \quad \dots (5.36) \end{aligned}$$

For elastic-perfectly plastic material

$$\frac{\partial b}{\partial \bar{e}^p} = 0 \quad \text{and} \quad \frac{\partial f_y}{\partial \epsilon_{ij}^p} = 0 \quad \text{and hence } H = 0$$

5.2.3.6 Undrained Elastic-Plastic Stress-Strain Relationship

For the solution of general boundary value problem under undrained condition, the total stress is expressed in terms of effective stress, σ'_{ij} , and pore pressure, p , from Eqn. (3.6e) as

$$\sigma_{ij} = \sigma'_{ij} + \delta_{ij} \alpha p \quad \dots (5.37)$$

Using Eqn. (5.35) and Eqn. (3.6e) under undrained condition, ($\xi = 0$) one can write,

$$\{\dot{\sigma}_{ij}\} = [D_{sd}^{Ep}] \{\dot{\epsilon}_{ij}\} + \alpha^2 M_o \{\dot{\epsilon}_{ij}\} \quad \dots (5.38)$$

where $[D_{sd}^{Ep}]$ is defined as solid skeleton constitutive matrix. It has been pointed out before that the constitutive matrix, $[D_{sd}^{Ep}]$, is responsible for all deformations due to material nonlinearity. The elasto-plastic form of the $[D_{sd}^{Ep}]$ matrix is given by Eqn. (5.38) and after some algebraic manipulation one can obtain (Appendix B)

$$\{\dot{\sigma}_{ij}\} = \begin{bmatrix} \lambda + 2\mu + \alpha^2 M_o - \frac{E_{11}^2}{L} & \lambda + \alpha^2 M_o - \frac{E_{11} \cdot E_{22}}{L} & \lambda - \frac{E_{11} \cdot E_{33}}{L} & - \frac{E_{11} \cdot E_{12}}{L} \\ & \lambda + 2\mu + \alpha^2 M_o - \frac{E_{22}^2}{L} & \lambda - \frac{E_{22} \cdot E_{33}}{L} & - \frac{E_{22} \cdot E_{12}}{L} \\ & & \lambda + 2\mu + \alpha^2 M_o - \frac{E_{33}^2}{L} & - \frac{E_{12} \cdot E_{33}}{L} \\ & & & \mu - \frac{E_{12}^2}{L} \end{bmatrix} \{\dot{\epsilon}_{ij}\}$$

SYM.

... (5.39)

where

$$a = \frac{\partial f_y}{\partial J_1}$$

$$B = \frac{\partial f_y}{\partial J_2} = \frac{\mu}{2\sqrt{J_2}}$$

$$L = 3a^2 (3\lambda + 2\mu) + \mu$$

$$E_{11} = a (3\lambda + 2\mu) + 2 BS_x$$

$$E_{12} = 2 B\tau_{xy}$$

$$E_{22} = a (3\lambda + 2\mu) + 2 BS_y$$

$$E_{33} = a (3\lambda + 2\mu) + 2 BS_z \quad \dots (5.40)$$

5.3 Solution for Equations of Motion

The general dynamic equations of motion of a gravity type offshore structure-foundation system resting on saturated soils can be written in a global assembled form as

$$[M^*] \{\ddot{\underline{U}}\} + [C^*] \{\dot{\underline{U}}\} + [K^*] \{\underline{U}\} = \{P_B\} \dots (5.41)$$

where, $[M^*]$, $[C^*]$ and $[K^*]$ are the assembled global mass, damping and stiffness matrices respectively. $\{P_B\}$ is the global load vector which takes into account the effect of wave loading.

For, linear systems, Eqn. (5.41) can be solved by any one of the following methods.

- a. Modal analysis or modal superposition.
- b. Direct integration.
- c. Fourier analysis or complex response.

In linear analysis the response can be computed either in the time domain or in the frequency domain, but for a true nonlinear analysis (geometric or material) time domain is the most suitable one.

5.3.1 Equations of Motion in Incremental Form

In the case of a nonlinear system, the incremental form of the equations of motion can be expressed in the following form for time, $t+\tau$, as:

$$[K^*] \{\Delta \underline{U}\} = \{P_B\}_{t+\tau} - \{\tilde{R}\}_t - [M^*] \{\ddot{\underline{U}}\}_{t+\tau} - [C^*] \{\dot{\underline{U}}\}_{t+\tau} \quad \dots (5.42a)$$

where

$$\{\Delta \underline{U}\} = \{\underline{U}\}_{t+\tau} - \{\underline{U}\}_t \quad \dots (5.42b)$$

= incremental displacement vector

$$R_t = \sum^M \int_V [\underline{\psi}]^T \{\sigma\}_t dV \quad \dots (5.42c)$$

= residual load vector computed from the stresses vector $(\sigma)_t$, at time t for all elements and $[\underline{\psi}]^T$ obtained from Eqn. (3.25).

and

τ = increment of time step.

Several methods are available for solving the above equations of motion using direct integration scheme. In this investigation, an implicit integration scheme is used where the solution is advanced by one time step interval, Δt . The values of the displacement and its derivatives at one instant of time is then used to determine the values at subsequent time steps by means of recurrence relationships.

5.3.1.1 Wilson - θ Method

In this method, the integration is advanced by first linearly interpolating to a hypothetical time step increment, $\tau = \theta \Delta t$, ($\theta = 1.37$) and thus obtaining the displacement and its derivatives at $\tau = \theta \Delta t$, time point. Later the displacement, and its derivatives at $t = t + \Delta t$ are computed by the averaging procedure.

The linear acceleration method assumes that the velocities and displacements are related to the acceleration in the following form:

$$\dot{\underline{U}}_{t+\tau} = \dot{\underline{U}}_t + \frac{\tau}{2} (\ddot{\underline{U}}_{t+\tau} + \ddot{\underline{U}}_t) \quad \dots (5.43a)$$

$$\underline{U}_{t+\tau} = \underline{U}_t + \tau \dot{\underline{U}}_t + \frac{\tau^2}{6} (\ddot{\underline{U}}_{t+\tau} + 2\ddot{\underline{U}}_t) \quad \dots (5.43b)$$

which gives

$$\underline{U}_{t+\tau} = \frac{6}{\tau} \Delta \underline{U} - \frac{6}{\tau} \dot{\underline{U}}_t - 2\ddot{\underline{U}}_t \quad \dots (5.43c)$$

and

$$\dot{\underline{U}}_{t+\tau} = \frac{3}{\tau} \Delta \underline{U} - 2\dot{\underline{U}}_t - \frac{\tau}{2} \ddot{\underline{U}}_t \quad \dots (5.43d)$$

Substituting Eqns. (5.43c) and (5.43d) in Eqn. (5.42a) one obtains

$$[\hat{\mathbf{K}}]_t \{\Delta \underline{U}\} = \{\hat{\mathbf{P}}_B\}_{t+\tau} \quad \dots (5.44)$$

where

$[\hat{\mathbf{K}}]$ = effective undrained stiffness matrix

$$= [\mathbf{K}^*]_t + \frac{6}{\tau} [\mathbf{M}^*]_{t+\tau} + \frac{3}{\tau} [\mathbf{C}^*]_{t+\tau} \quad \dots (5.44a)$$

and

$$\begin{aligned}
 \{\hat{P}_B\}_{t+\tau} &= \text{effective load vector} \\
 &= \{P_B\}_{t+\tau} + [M^*] \left(\frac{6}{\tau} \dot{U}_t + 2\ddot{U}_t \right) + [C^*] \\
 &\quad \left(2\dot{U}_t + \frac{\tau}{2} \ddot{U}_t \right) - \{\tilde{R}\}_t \quad \dots (5.44b)
 \end{aligned}$$

5.3.2 Incremental Analysis

In an elasto-plastic analysis, the stress-strain relationship is generally formulated in an incremental form. The basis of the incremental formulation is to subdivide the real loading function into a finite number of small increments which are then prescribed one at a time. During the load increment, the stiffness matrix remains constant for that loading step. As the load is increased, the foundation reaches its elastic limit when the highest stressed element begins to yield following a specific yield model. Once the material has yielded, the stress-strain relationship can only be defined in terms of existing state of stress and can only be evaluated in an incremental form. The stiffness matrix is recomputed at every load step by substituting the appropriate elastic-plastic

stiffness matrices for yielded elements. During each time of load step increment, an iterative approach in combination with the well known modified Newton-Raphson (MNR) technique is used with the total applied load subdivided into a finite number of load increments. For a dynamic analysis this means the evaluation of the effective load vector at each time step.

For each load/time increment, the incremental deflection is determined using the modified Newton-Raphson technique. If the solution at the start of a load increment is known, the tangent stiffness matrix at that point can be computed and used to obtain an initial linear estimate of the deflection at the end of the increment. However, since the load-deflection or stress-strain curve of the soil is non-linear, the actual load the system can support at this deflection is less than the applied load and thus a load unbalance exists at the end of the first iteration. The unbalanced loads are applied to the system and a revised estimate of the incremental deflection is determined using the same tangent stiffness matrix evaluated at the previous point. This process is repeated until a saw-tooth type convergence is achieved.

5.3.2.1 Iteration of Equilibrium Equations

Eqn. (5.44) was obtained by linearizing the system response about the configuration at time $t = t + \tau$. It may be noted here that the errors in the linearization can be expected to be small provided the load increments or time steps are very small. As the time step increment becomes very small, the cost of the analysis will increase tremendously and thus it is quite often necessary to imply larger time step increment for dynamic analysis or load step increment for static analysis. This large increment can introduce gross instability by slowly drifting away the computed solution from the exact solution. This is much more pronounced in the dynamic analysis than in the static analysis, since in the dynamic analysis the solution is always dependent on the past history. Therefore, in order to obtain a reasonable accurate solution, equilibrium iteration is necessary within a load and/or time step with a required tolerance criteria.

The basic equations for equilibrium iteration are obtained by using modified Newton-Raphson (MNR) technique and are given in matrix form as

$$[\hat{K}]_t^{(i-1)} \{\Delta \underline{U}\}^{(i)} = \{P_B\}_{t+\tau}^{(i-1)} \quad \dots (5.45a)$$

where

$$\{\Delta \underline{U}\}^{(i)} = \{\underline{U}\}_{t+\tau}^{(i)} - \{\underline{U}\}_{t+\tau}^{(i-1)} \quad \dots (5.45b)$$

and i = number of iterations within a particular time or/load step.

Eqn. (5.45a) forms the basis for the iterative solution of Eqn. (5.44) with the following initial conditons as

$$[\hat{K}]_{t+\tau}^{(0)} = [\hat{K}]_t, \quad \{\tilde{R}\}_{t+\tau}^{(0)} = \{\tilde{R}\}_t \quad \text{and} \quad \{\Delta \underline{U}\}_{t+\tau}^{(0)} = \{\Delta \underline{U}\}$$

... (5.46)

The iteration procedure is continued until a suitable convergence has been achieved in the solution process during a particular time/load step increment. The suitable convergence criteria is obtained by specifying a fixed tolerance (RTOL) at the beginning of the solution and by checking the ratio of the norm of the increment of displacement to the total displacement as

$$\left\| \frac{\Delta \underline{U}^{(i)}}{\underline{U}^{(i)} + \underline{U}_t} \right\|_2 < \text{RTOL} \quad \dots (5.47)$$

5.4 Permanent Deformation of A Strip Footing

The incremental elastic-plastic analysis is used here to compute the permanent shakedown displacement of a strip footing under cyclic vertical loading condition. The analysis is carried out for a first few cycles of the loading until a steady-state condition is reached.

The simple finite element model of a strip footing underlaid by a shallow stratum of clay under undrained condition is shown in Fig. 5.6. The mesh consists of 32 elements and 45 nodes. The ratio of the footing width to the depth of the stratum is 0.80. Because of the symmetry of the vertical loading, the mesh for half of the footing is shown in Fig. 5.6. The loading diagram is similar to a triangular pulse with a period of 1 second. The loading magnitude is defined by a dimensionless parameter, F , equal to the ratio of weight per unit area to the undrained shear strength (i.e. $F = W/As_u$ in which W is the vertical load, A = area of the strip footing and s_u = undrained shear strength of the soil). Five cycles of loading (5 seconds duration) are considered in the analysis and a total of eighty (80) load steps are used in order to trace the entire load path.

Vertical side boundary is placed at a distance of

MESH GENERATION

NODAL POINT NUMBERING

ELEMENT NUMBERING

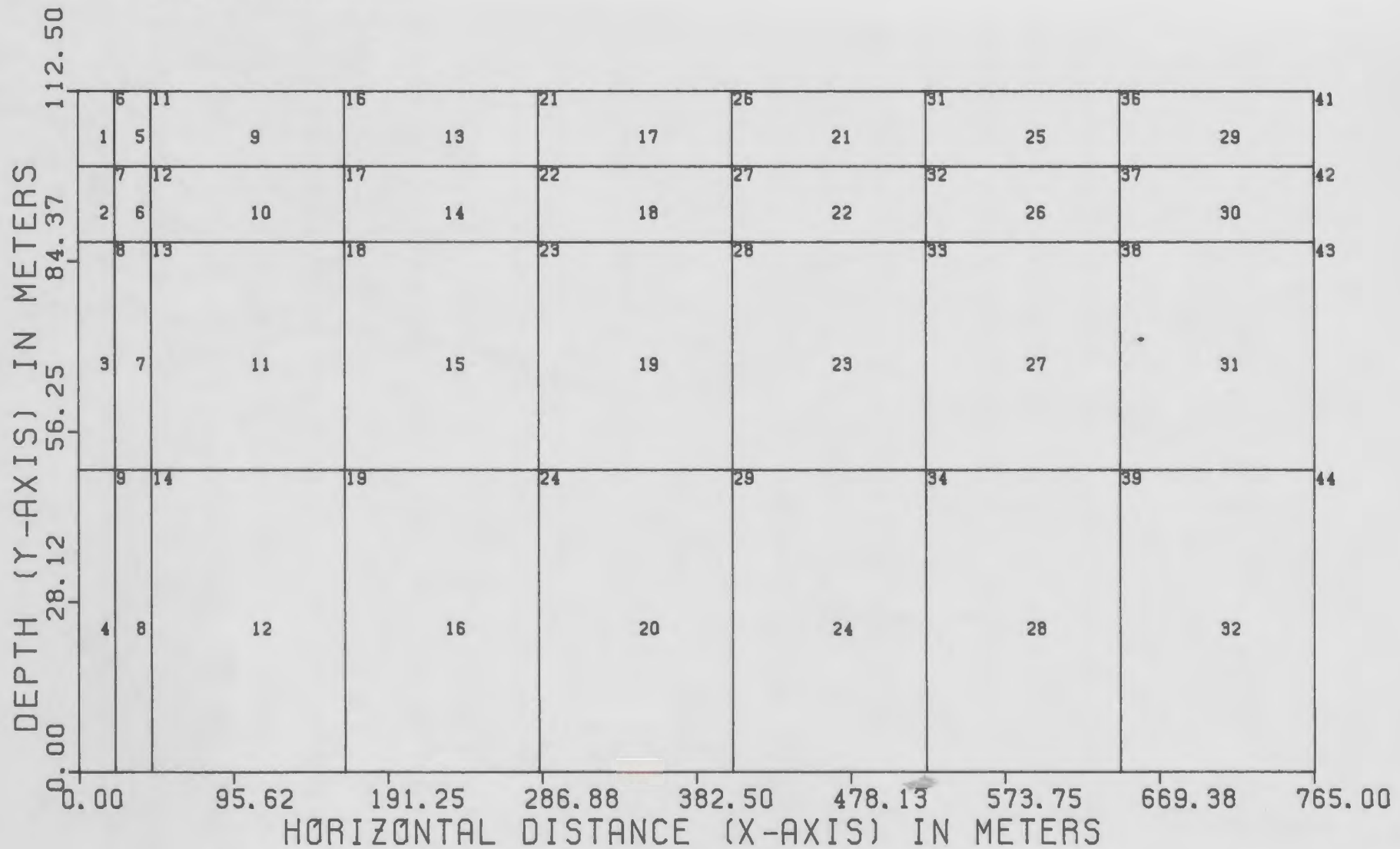


FIG. 5.6 FINITE ELEMENT MESH FOR STRIP FOUNDATION
(INCREMENTAL ANALYSIS)

eight and a half times the width of the footing and assumed to be smooth. The bottom boundary is assumed to be rough and rigid. The footing is assumed to deform under plane strain condition. Reduced integration technique is used to ensure the nearly incompressibility of the undrained soil medium.

In the elastic-plastic undrained analysis the soil is modelled as a weightless two-phase material with $s_u = 100$ KPa and $\phi_u = 0$, where s_u = undrained shear strength and ϕ_u = angle of internal friction. The drained elastic Lamé's parameters $\lambda = 150 s_u$, $\mu = 100 s_u$ and drained Poisson's ratio, $\nu_d = 0.30$ are assumed for the soil skeleton whereas the bulk modulus for pore water is assigned as $K_w = 2000$ MPa. This gives an equivalent undrained Poisson's ratio, $\nu_u = 0.499$ approximately.

In order to compute the permanent deformation of the strip footing, the elastic and shakedown limit pressures obtained in Chapter 4 (Refer Table 4.2) are used here. It is found that for $F^e < 3.28$ (dimensionless elastic limit pressure) the behaviour of the soil foundation is purely elastic. However, for $F = F^s = 5.31$, the foundation undergoes cyclic inelastic deformation for few cycles and then reaches a steady state situation. The reason for applying a shakedown

pressure of $F^S = 5.31$ was to obtain the boundedness of the deformation. Table 5.2 shows the elastic and inelastic deformation for the first five cycles of loading. For $F^S = 5.31$, the permanent (inelastic) deformation increases for the first two cycles of loading and then remains unchanged implying that the plastic work is bounded in time and a limited amount of plastic flow has occurred. The foundation develops residual stresses due to the initial plastic flow and therefore the stresses obtained from the subsequent cyclic elastic-plastic analysis can be superimposed on these residual stresses without violating the yield criterion (shakedown condition). Thus, the foundation shakes down to elastic behaviour and any load $3.28 \leq F^S < 5.31$ will be a safe load in order to achieve a shakedown condition. The inelastic (permanent) deformation computed for the footing is 0.82% of the footing width and shown in Table 5.2.

TABLE 5.2 - PERMANENT DEFORMATION OF A STRIP FOOTING
SUBJECTED TO REPEATED VERTICAL LOADING

Cycle Number	(Vertical Displacement/Footing Width) in Percent		
	Elastic	Permanent	Total
1/4	0.38	-	0.38
1/2	0.76	0.79	1.56
3/4	0.38	0.81	1.19
1	0.00	0.81	0.81
1 1/4	0.38	0.81	1.19
1 1/2	0.76	0.82	1.58
1 3/4	0.38	0.82	1.20
2	0.00	0.82	0.82
2 1/4	0.38	0.82	1.20
2 1/2	0.76	0.82	1.58
2 3/4	0.38	0.82	1.20
3	0.00	0.82	0.82
3 1/4	0.38	0.82	1.20
3 1/2	0.76	0.82	1.58
3 3/4	0.38	0.82	1.20
4	0.00	0.82	0.82
4 1/4	0.38	0.82	1.20
4 1/2	0.76	0.82	1.58
4 3/4	0.38	0.82	1.20
5	0.00	0.82	0.82

CHAPTER VI

RESPONSE ANALYSES OF A GRAVITY TYPE
OFFSHORE FOUNDATION CONSIDERING SOIL SHAKEDOWN

6.1 Introduction

Three types of analyses are carried out in order to compute the permanent deformations beneath a gravity type offshore foundation. The first analysis investigates the effects of eccentricity and angle of inclination of the load on the shakedown load factor for a flexible foundation.

In the second analysis, the permanent deformation of the foundation of a gravity type offshore platform is studied considering the static soil shakedown. In the third analysis the dynamic shakedown load factor is first determined and the responses are then computed and compared for various types of soil modelling. In the second and third analyses the effect of the caisson structure is taken into account by modelling it as a stiff footing.

6.2 Method of Approach

The above analyses are carried out using the two-dimensional finite element plane strain formulation described in Chapters 3, 4 and 5. The detailed descriptions of the computer programme especially developed for the present investigation and the necessary input data requirements are presented in Appendices C and D.

6.2.1 Limitations of the Method

The limitations of the present approach are:

- a. the effect of the change in volume due to shear deformation is neglected.
- b. the soil beneath the foundation is assumed to be linearly elastic-perfectly plastic material and is modelled with the Von Mises yield criteria following associated flow theory of plasticity. The assumption of constant soil shear strength which is independent of the normal stress components is a valid approximation for undrained saturated clay, the so called $\phi_u = 0$ condition (Skempton, 1941).

- c. Present analyses are based on small strain formulation i.e. they do not consider the non-linearity caused by the change in geometry.

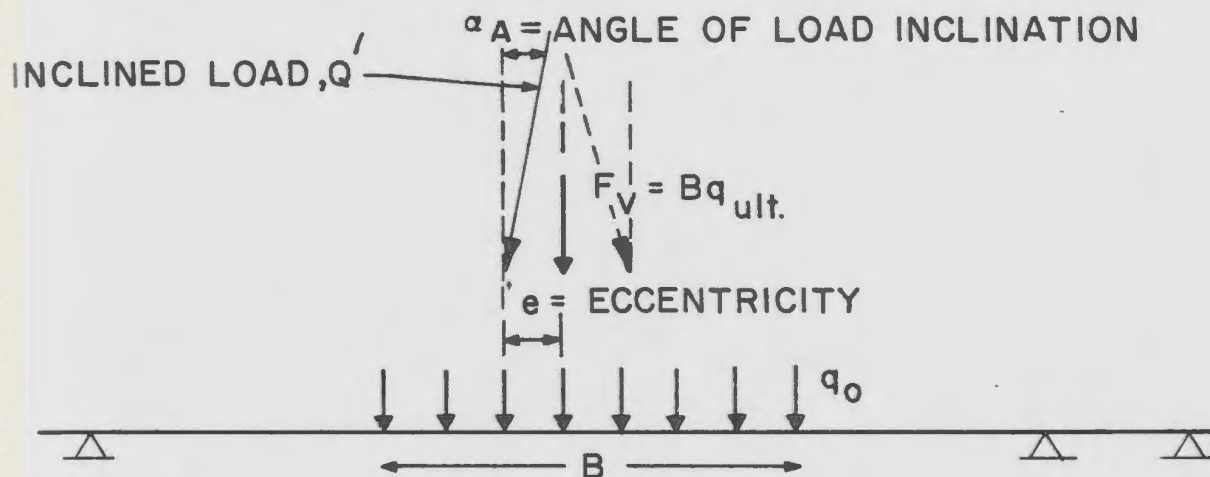
6.3 Flexible Foundation Analysis

Several analyses are carried out in order to determine the effects of the eccentricity and the angle of inclination of the cyclic wave loading on the static elastic and shakedown limit pressures.

6.3.1 Geometry and Finite Element Mesh

Fig. 6.1 shows the geometry of the foundation resting on saturated cohesive soil. Based on the bearing capacity formula, two load inclination angles, α_A , and five ratios of eccentricity to footing width, e/B , have been chosen for computing the elastic and shakedown limit pressures. Von Mises yield condition is used in all analyses and represented by six linearized yield planes.

Fig. 6.2 depicts the finite element mesh for the flexible strip foundation. The foundation has a loaded width of ninety meters. The soil medium is assumed to be homogeneous and isotropic. The



FOOTING WIDTH, $B = 90.0 \text{ m}$

UNDRAINED SHEAR STRENGTH, $S_u = 100.0 \text{ kN/m}^2$

BULK WEIGHT, $\rho = 18.6 \text{ kN/m}^3$

SPECIFIC WEIGHT OF WATER, $\rho_f = 9.8 \text{ kN/m}^3$

$$q_{ult} = \frac{F_V}{B'L} = N_c \times S_u \times \left(1 - \frac{\alpha_A}{90}\right)^2 \left(1 - 2 \frac{e}{B}\right)$$

$$\text{WHERE } \alpha = \tan^{-1}\left(\frac{F_H}{F_V}\right)$$

$$N_c = (2 + \pi)$$

B' = EFFECTIVE WIDTH

FIG. 6.1 BEARING CAPACITY FOR FLEXIBLE FOOTING ON COHESIVE SOIL (MEYERHOF, 1963)

MESH GENERATION

NODAL POINT NUMBERING
ELEMENT NUMBERING

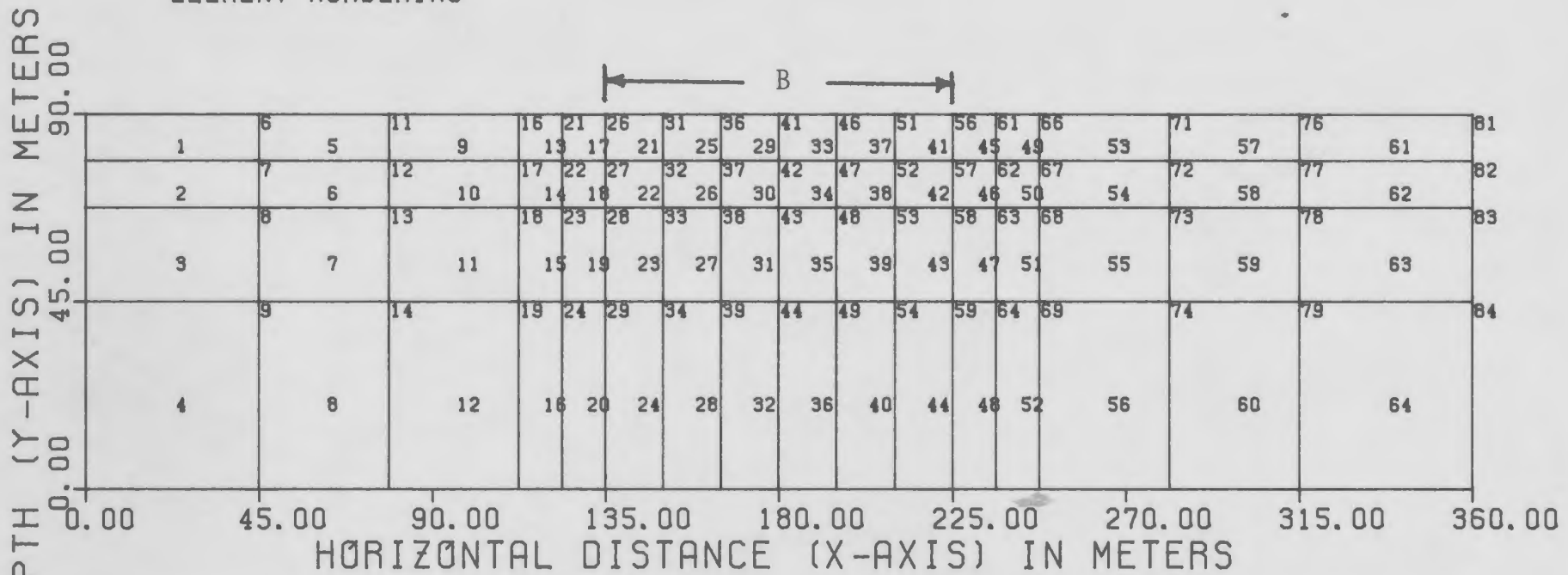


FIG. 6.2 FINITE ELEMENT MESH FOR STRIP FOUNDATION

mesh consists of 85 nodes and 64 four noded rectangular isoparametric elements. Vertical side boundary is placed at a distance two times the width of the footing from the footing centerline and assumed to be smooth. The bottom boundary is assumed to be rough and rigid. The footing is assumed to deform under plane strain condition. Reduced integration technique is used to implement effectively the nearly incompressibility condition for undrained deformation.

In the undrained analysis the soil is modelled as a weightless two-phase material with $s_u = 100.0$ kPa and $\phi_u = 0.0$. The drained elastic Lamé's parameters $\lambda = 150 s_u$, $\mu = 100 s_u$ and drained Poisson's ratio, $\nu_d = 0.30$ are assumed for the soil skeleton whereas the bulk modulus for pore water is assigned as $K_w = 2000$ MPa. This gives an equivalent undrained Poisson's ratio, $\nu_u \approx 0.499$.

The submerged weight, F_V , of the gravity type offshore structure is assumed to act with an eccentricity, $\pm e$, which causes a cyclic moment, $F_M = F_V \cdot (\pm e)$ at the base level (Fig. 6.1). The foundation is also subjected to the horizontal force component, $\pm F_H$, of the cyclic inclined

loading, Q' , as shown in Fig. 6.1. The stress distributions due to loadings, F_V , F_H and F_M are assumed to be uniform and linear respectively.

6.3.2 Evaluation of Elastic Limit Load

For the vertical load alone ($e/B = 0.0$, $\alpha_A = 0.0$), the vertical pressure is applied at each of the seven nodes 26, 31, 36, 41, 46, 51, and 56.

The stress fields comprising of σ_x , σ_y , σ_z and τ_{xy} for all the elements are obtained by applying the vertical pressure and by observing the pressure at which one/or more of the elements will first become plastic.

The elastic limit load factor, β_E is obtained from Eqn. (5.25d) and Eqn. (5.26b) as

$$\beta_E = \frac{s_u}{\sqrt{\frac{1}{2}(s_x^2 + s_y^2 + s_z^2) + \tau_{xy}^2}} \quad \dots(6.1)$$

and computed for each element. From the elastic half space theory the elastic limit pressure is given as $3.14 s_u$ (Jumikis, 1969). This pressure is obtained based on Tresca failure surface. However, the present analysis indicates a value of

3.08 s_u which is within the 2.0 % of the theoretical elastic limit pressure solution. Element No. 31, at a depth of 34.0 meters from the surface indicates a minimum load factor, β_E , of 1.29. The vertical displacement at center node 41, is compared with that obtained based on Kelvin's equation for a line load acting within an infinite solid (Taylor and Matyas; 1983).

For a clay layer with constant Young's modulus the vertical settlement may be estimated following Taylor and Matyas (1983) as

$$\delta_V = \frac{\sigma_V d_o}{\pi u E} \cdot \frac{1 + v_u}{1 - v_u} \left[\alpha_o + (1 - 2v_u) \alpha_1 \right] \quad (\dots(6.2))$$

where

δ_V = vertical settlement in meter

σ_V = applied vertical pressure = 233 kPa

d_o = depth of the foundation = 90 m

B = foundation width = 90 m (Fig. 6.1 and Fig. 6.2)

v_u = Poisson's ratio = 0.499

$E = \text{Young's modulus} = 29,980 \text{ kPa}$

and α_0 and α_1 are taken from Fig. 5 of the above reference; for $d_0/B = 1.0$, $\alpha_0 = 0.3571$ and $\alpha_1 = 2.5833$.

Based on the above parameters, δ_V is computed as 0.2415 meter as opposed to the vertical displacement of 0.2120 meter obtained from the finite element analysis. The difference in the result can be explained due to the effect of the side boundary.

6.3.3 Evaluation of Shakedown Limit Load

The computer programme has been checked in Chapter 4 for a strip footing with the available solutions under vertical loading condition. In this section, the static shakedown analysis of a flexible foundation is carried out under cyclic inclined eccentric loading condition. Fig. 6.1 depicts the loading system and notation.

The piecewise linearized Von Mises yield criterion with its associative nature of flow rule condition is used in shakedown analyses. The finite element

mesh of the soil stratum is shown in Fig. 6.2 and the material properties are described in Section 6.3.1.

For the case of centered vertical loading (i.e.

$\alpha_A = 0.0$ and $e/B = 0.0$), the shakedown limit pressure, q_s , is obtained as $5.22 s_u$ from Eqn. (4.55) and shown in Table 6.1. The above pressure is 1.5% higher than the true collapse pressure. The typical central processing unit (cpu) time required is 180 minutes and major portion of this time is spent in solving the linear programming problem involving 193 constraints and 640 variables.

6.3.4 Influence of Eccentricity and Load Inclination Angle

The parametric studies are carried out in order to determine the influence of the various, e/B , ratios and the angle of inclination of the cyclic wave loading, α_A , on the elastic and shakedown limit pressures.

The elastic limit pressures, q_e , for eccentricity ratios of $e/B = 0.05$ to 0.16 are also computed at increments of 0.05 for $\alpha_A = 0$ and shown in

TABLE 6.1 - COMPARISON OF ELASTIC AND SHAKEDOWN LIMIT PRESSURES FOR $e/B = 0.0 - 0.16$ AND $\alpha_A = 0.0^\circ$

e/B	Number of elements	Number of Nodal Points	Number of Equations	Elastic Limit Pressure (q_e/s_u)	Shakedown Limit Pressure (q_s/s_u)
0.0	64	85	128	3.08	5.22
0.05	64	85	128	2.75	4.10
0.10	64	85	128	2.28	3.60
0.15	64	85	128	2.10	3.15
0.16	64	85	128	1.98	3.10

Table 6.1. The elastic limit pressure, q_e , is reduced as much as 35% for an eccentricity ratio $e/B = 0.16$ as compared to that corresponding to $e/B = 0.0$. For $\alpha_A = 10^\circ$, the uniform horizontal pressure is also applied in addition to the distributed vertical pressure to the nodes at the top of the caisson. The elastic limit pressure, q_e , for $e/B = 0.0$ is computed as $2.98s_u$. In this case, the minimum value of β_E , is obtained in element No. 35. Table 6.2 illustrates the values of the elastic limit pressure, q_e , for all e/B ratios.

For $\alpha_A = 0.0$, Table 6.1 compares the values of the static shakedown limit pressures for various e/B ratios from 0.0 to 0.16, whereas Table 6.2 compares the same for $\alpha_A = 10.0^\circ$. Results show that the shakedown limit pressures decrease as the e/B ratio and the inclination angle increase. The reduction could be as high as 54% e.g. for $e/B = 0.16$ and $\alpha_A = 10^\circ$ as opposed to $e/B = 0.0$ and $\alpha_A = 0.0$. Figs. 6.3(a) and (b) depict the comparison of these values with those obtained from semi-empirical bearing capacity relationship (Fig. 6.1) given by Meyerhof (1963). The shakedown pressures are obtained from an upperbound solution and therefore the use of the semi-empirical formula will

TABLE 6.2 - COMPARISON OF ELASTIC AND SHAKEDOWN LIMIT PRESSURES FOR $e/B = 0.0 - 0.16$ AND $\alpha_A = 10.0^\circ$

e/B	Number of elements	Number of Nodal Points	Number of Equations	Elastic Limit Pressure (q_e/s_u)	Shakedown Limit Pressure (q_s/s_u)
0.0	64	85	128	2.98	3.92
0.05	64	85	128	2.63	3.52
0.10	64	85	128	2.23	3.00
0.15	64	85	128	1.90	2.50
0.16	64	85	128	1.80	2.40

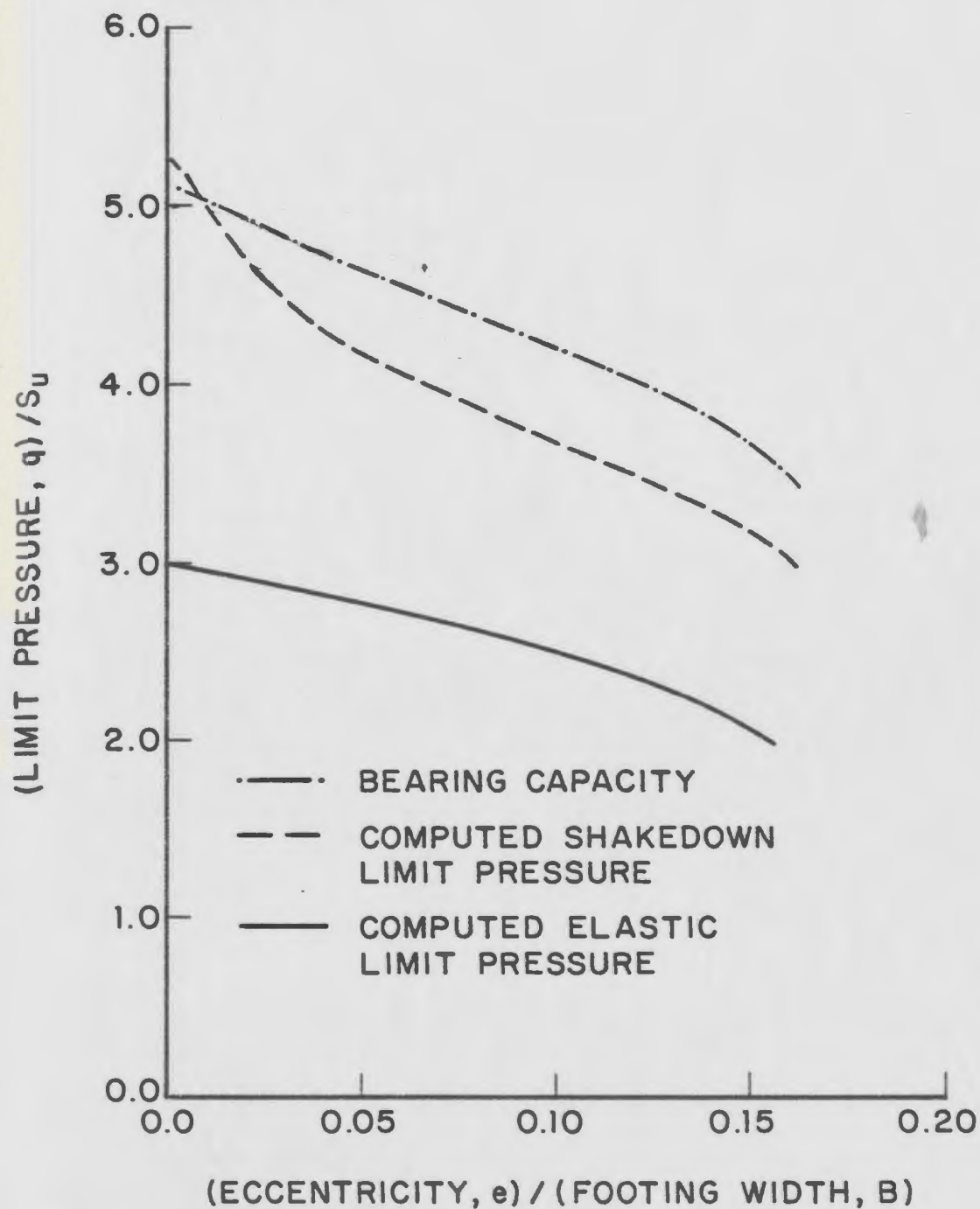


FIG. 6.3 (a) ELASTIC AND SHAKEDOWN LIMIT PRESSURES FOR VARIOUS ECCENTRICITIES ($\alpha_A = 0.0^\circ$)

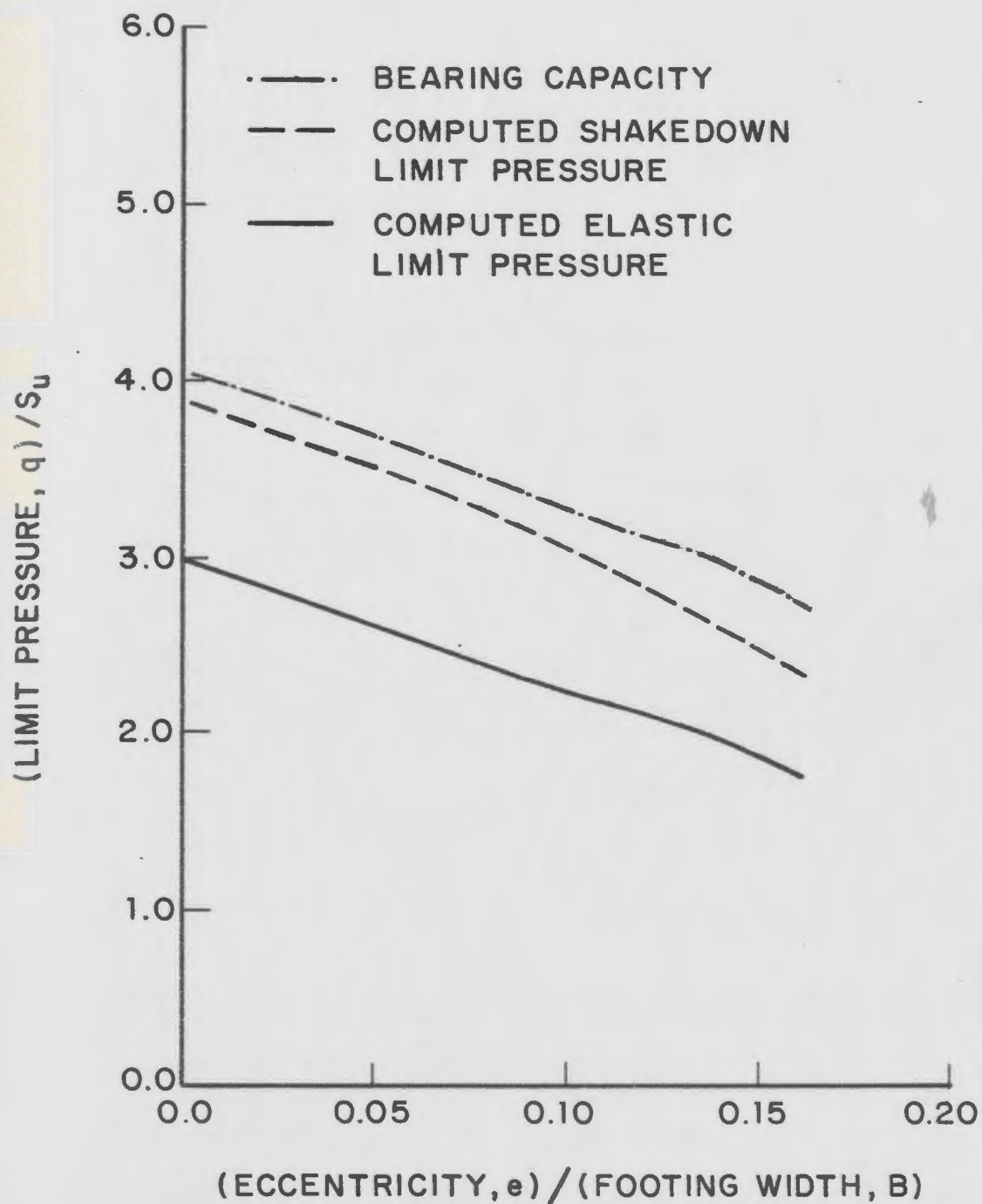


FIG. 6.3 (b) ELASTIC AND SHAKEDOWN LIMIT PRESSURES FOR VARIOUS ECCENTRICITIES ($\alpha_A = 10^\circ$)

not produce a conservative estimate of the bearing capacity of the foundation when the loading is repetitive in nature.

6.4

Static Analysis of a Gravity Type Offshore Foundation

In this section the formulation presented in the previous chapters is used to analyze the action of an offshore gravity structure with its soil foundation which consists of saturated clays resting on a rigid base. Particular attention is given to the static nature of the wave loading and the soil foundation is assumed to behave in an undrained manner during one particular storm. The reduced integration technique is used in order to implement effectively the nearly incompressibility condition for undrained deformation.

The permanent deformation of the caisson structure is computed by using an elastic-plastic finite element formulation as described in Section 5.2.3. A static shakedown analysis is first carried out for the entire caisson-foundation-soil system. Having computed the static shakedown load factor, the loading domain at shakedown is first established and the responses are then computed and

compared for two types of soil behaviour: a) linear and b) nonlinear.

6.4.1 Loading Programme

The magnitude of the wave loading on the structure-foundation depends on the wave period, T_i , length, L_i , and height, H_i and the dimensions of the structure. In a typical storm situation there are varying number of waves having similar characteristics and can be grouped together. However, it has been observed that the complete storm in a deep ocean environment consists of wave groups having smaller heights followed by wave groups having larger heights until the peak of the storm has been reached. In the present analysis a single wave with the largest height with maximum return period has been considered. The individual wave has a wave length of 218.0 meters, height 30.0 meters and period 12.0 seconds. Before the storm, structure is in calm water and as such the total vertical force, F_V , acting on the structure is the buoyant (submerged) weight of the structure.

Fig. 6.4(a) shows a wave profile in the free field and Fig. 6.4(b) shows the structure standing in

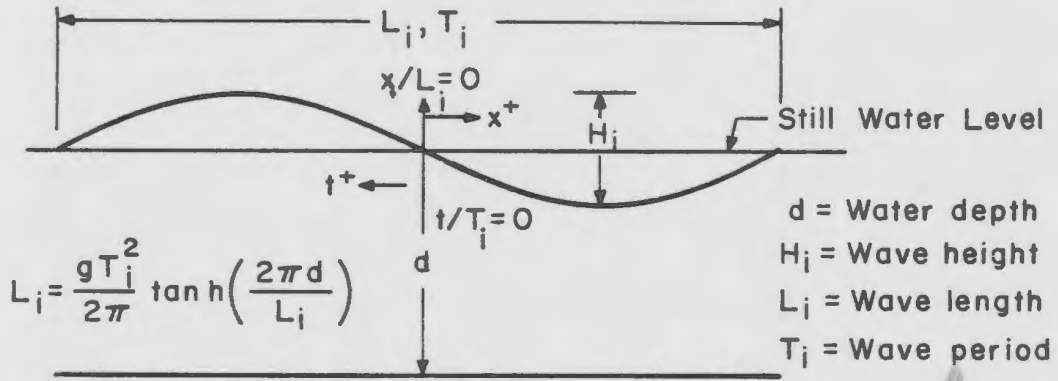


FIG. 6.4(a) CHARACTERISTICS OF A LINEAR WAVE

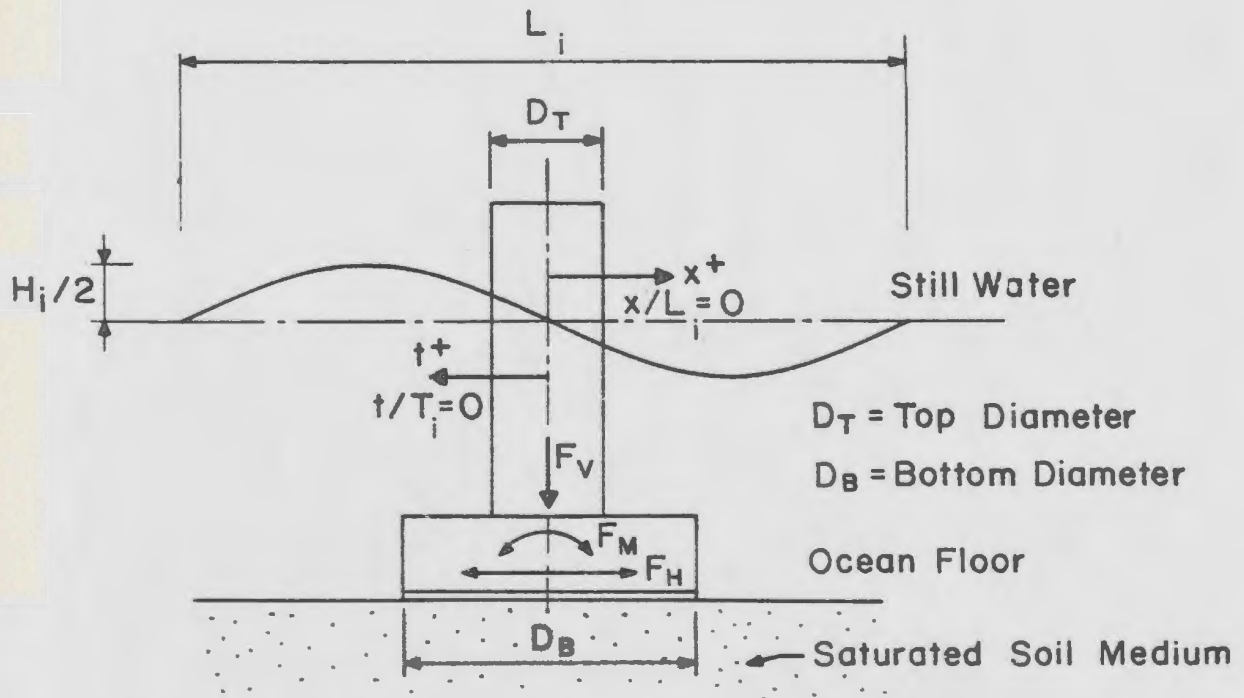


FIG. 6.4(b) WAVE FORCES ON STRUCTURE

water during a storm with its axis perpendicular to the line of the wave propagation. If the lateral dimension of the caisson is comparable to the wave length, the wave forces are primarily due to diffraction effects i.e. including inertial forces. The drag force is comparatively small and therefore can be neglected. The forces acting on the caisson due to a single wave is computed from diffraction theory and the values obtained have been taken from a problem given by Pool (1976). The net effect of the wave force is shown in Fig. 6.4(c) and represented by a cyclic horizontal force, $\pm F_H$, and a cyclic moment, $\pm F_M$. The general cyclic wave loading domain is therefore expressed as

$$\begin{aligned} & -\beta \bar{F}_H \leq F_H \leq \beta \bar{F}_H \\ \text{and} & \quad -\beta \bar{F}_M \leq F_M \leq \beta \bar{F}_M \end{aligned}$$

where \bar{F}_H and \bar{F}_M are the nondimensional parameters and defined as

$$\bar{F}_H = \frac{F_H}{A s_u}, \quad \text{and} \quad \bar{F}_M = F_H \cdot a_0;$$

where, β is the appropriate load factor, A is the area of the foundation under plane strain condition, a_0 is the height of the caisson structure

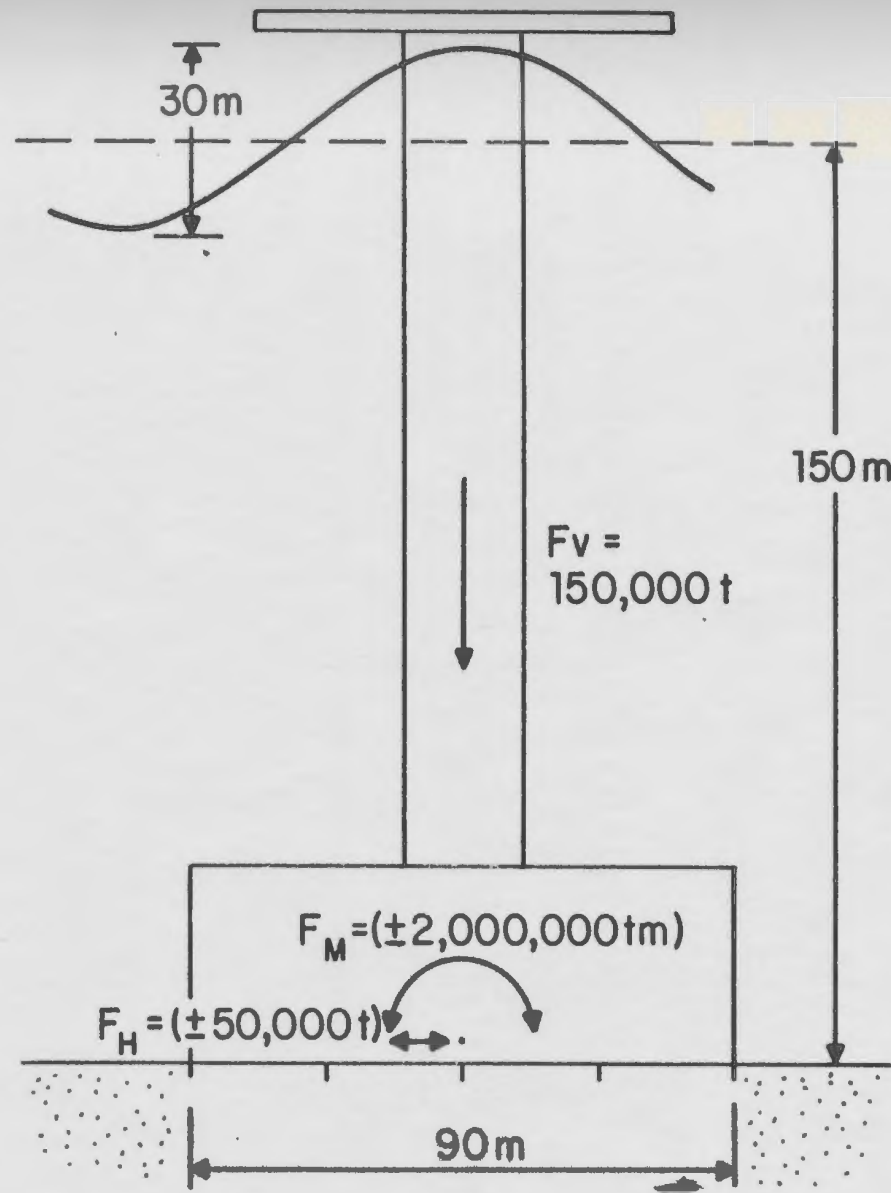


FIG. 6.4(c) TYPICAL LOADS ON A GRAVITY STRUCTURE (POOL, 1976)

above the mud line ($a_0 = 40.0$ meters) and \bar{F}_H and \bar{F}_M are prescribed values of F_H and F_M .

6.4.2 Modelling of the Caisson Structure and the Foundation Soil

The circular platform base shown in Fig. 6.4(c) has a diameter of 90 meters and is modelled as an equivalent rectangle with the same area. The rectangular base is later idealized as a plane strain model with the width equal to that of the structure. The total wave force is distributed per unit meter width of the structure. A two dimensional plane strain finite element model is used in carrying out the analysis. The caisson structure is modelled by two-dimensional four noded linear element whereas the foundation is modelled by the same type of elements using linear and nonlinear soil properties representing the characteristics of saturated cohesive soil.

The finite element mesh of the caisson structure-foundation system is shown in Fig. 6.5. The mesh consists of five structural elements (elements 21, 26, 31, 36, and 41) and sixty soil elements. The shear strength is assumed to be constant over the depth and the behaviour of the soil is assumed to

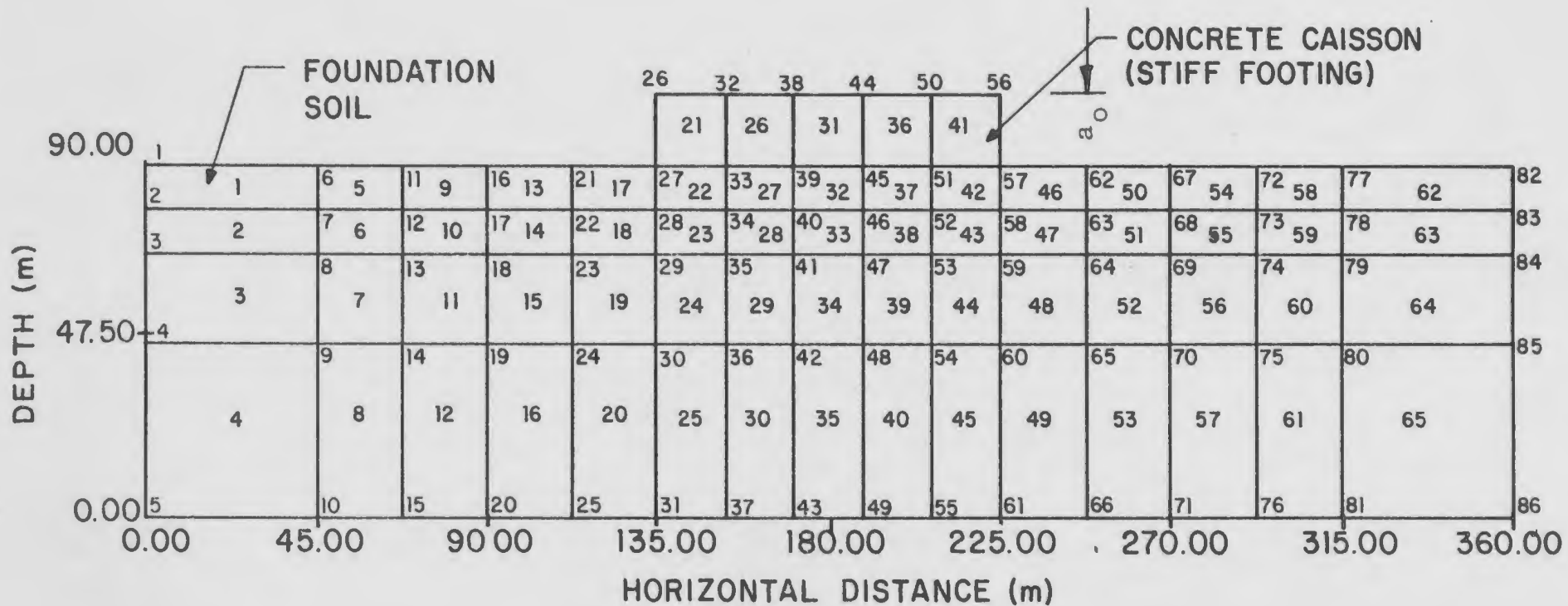


FIG. 6.5 FINITE ELEMENT MESH FOR STIFF FOUNDATION (NOT TO SCALE)

be undrained in nature with respect to the cyclic loading condition i.e. there is no flow of fluid with respect to soil.

In order to model the caisson foundation a very high stiffness is assigned to the structural elements 21, 26, 31, 36 and 41. The loads are applied to the foundation through the nodes 26, 32, 38, 44, 50 and 56. The stiffness ratio of the structure to the foundation soil, t_s , is computed from the following equation (Jumikis, 1969) as

$$t_s = \frac{1.25 E_L B^3}{E_f a_o^3} \quad \dots(6.3)$$

where E_L = Young's modulus of the soil in the
finite layer

E_f = Young's modulus of the footing

a_o = thickness of the footing

$B/2$ = half width of the footing

The semi-empirical method for obtaining the settlement at the center of the footing is given by Jumikis (1969) as follows:

$$\delta_V = [K(t) + 2K_r] \frac{\sigma_V B (1 - \nu_u)^2}{2\pi E_L} \dots (6.4)$$

where $K(t) = 0$ for $t_s = 0.0$ implying the rigid footing condition. The values of K_r for different, $\frac{2d_o}{B}$, ratios and variation of $K(t)$ with respect to t_s have been given by Jumikis (1969).

In the undrained analysis, the soil is treated as a weightless two phase material with $s_u = 100.0$ kPa, and $\phi_u = 0.0$ throughout the depth of the layer. The drained elastic parameters $\lambda = 150 s_u$, $\mu = 100 s_u$ and $\nu_d = 0.30$ are assumed for the soil skeleton and a high value of bulk modulus, 2000 MPa is assigned to the pore water.

The ratio of the Young's modulus for the caisson foundation to the soil medium is assumed as $\frac{E_f}{E_L} = 1000$. Therefore the stiffness ratio, t_s computed from Eqn. (6.3) becomes 0.0142, where $B/2 = 45.0$ meters and $a_o = 90.0$ meters. Because t_s is relatively small, it is reasonable to assume that the footing is stiff compared to the foundation soil. The computed vertical displacement at the center of the caisson is compared with that given by Jumikis (1969) and found to be in agreement. It is interesting to note that the vertical displacement of the stiff footing is found to be 25%

less than that compared with the flexible one.

6.4.3 Evaluation of Elastic Limit Load

The stress fields comprising of σ_x , σ_y , σ_z and τ_{xy} for all the elements are obtained by applying the cyclic horizontal loading, \bar{F}_H , ($\bar{F}_H = \frac{F_H}{As_u} = 0.78$) coupled with the vertical submerged weight of the structure, $\frac{F_V}{As_u} = 2.33$, at nodes 26, 32, 38, 44, 50 and 56. The elastic limit load factor, β_E , is then computed from Eqn. (6.1) which translates into the following elastic loading domain; $\frac{F_V}{As_u} = 2.33$ and $\frac{F_H}{As_u} = 0.50$.

6.4.4 Evaluation of Shakedown Limit Load

In this section, the static shakedown analysis of a caisson-structure foundation system is carried out under cyclic inclined eccentric loading condition. The cyclic horizontal loading due to the wave action and the vertical submerged weight of the structure are applied at nodes which are located at the top of the caisson structure. The upper bound value of 1.28 for shakedown load factor is obtained from Eqn. (4.55). Therefore shakedown domain is described as follows; $\frac{F_V}{As_u} = 2.33$ and $+\frac{F_H}{As_u} = 0.64$. The total number of constraints and

variables for this problem are 196 and 654 respectively. The central processing unit (cpu) time for this problem is 210 minutes and the major portion of this time is again utilised for solving the linear programming problem.

6.4.5 Elastic Analysis

The linear stress analysis of the structure-foundation system in calm water is carried out with a net submerged vertical force, F_V . The analysis is based on the assumption of undrained soil behaviour and the contour plots for normal stress σ_y , and pore pressure, p , are shown in Figs. 6.6(a) and 6.6(b). Assuming the cyclic horizontal force is at its maximum value (i.e. wave phase angle is either 90° or 270° refer Fig. 1.1) the elastic stress analysis is also carried out with the cyclic horizontal force, F_H , coupled with the submerged weight, F_V , acting at the top nodes of the caisson structure.

The distributions of horizontal and vertical displacements at selected nodes along the horizontal direction are shown in Figs. 6.7(a) and 6.7(b). In this case, the horizontal force is assumed to act from left to the right of the

(SIGMA-Y) / (UNDRAINED SHEAR STRENGTH)

CONTOUR INTERVAL = 0.200

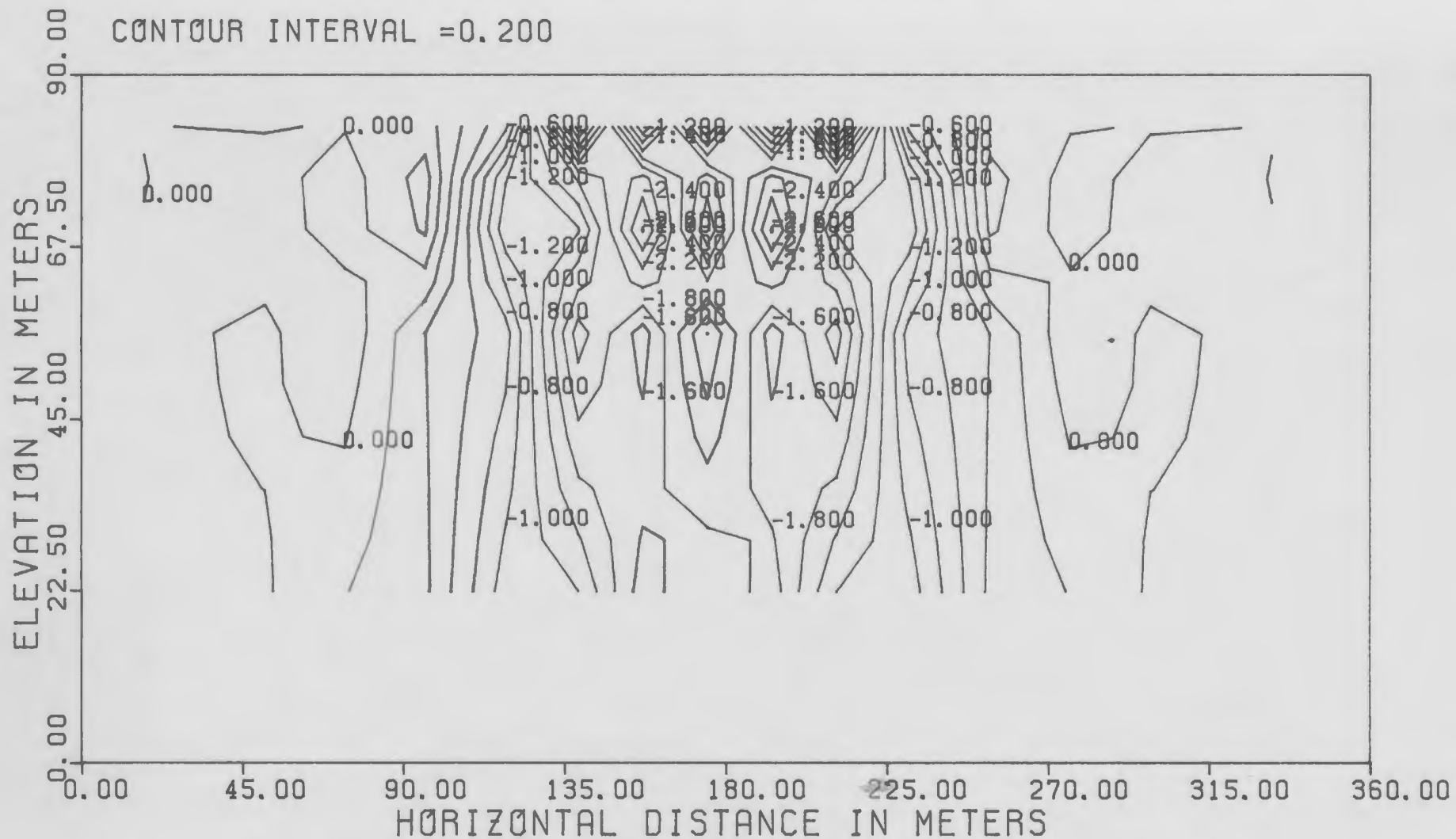


FIG. 6.6 (a) STRESS CONTOURS PLOT FOR (SIGMA-Y)
(WITH CAISSON; SUBMERGED WEIGHT ONLY)

(PORE PRESSURE) / (UNDRAINED SHEAR STRENGTH)

CONTOUR INTERVAL = 0.200

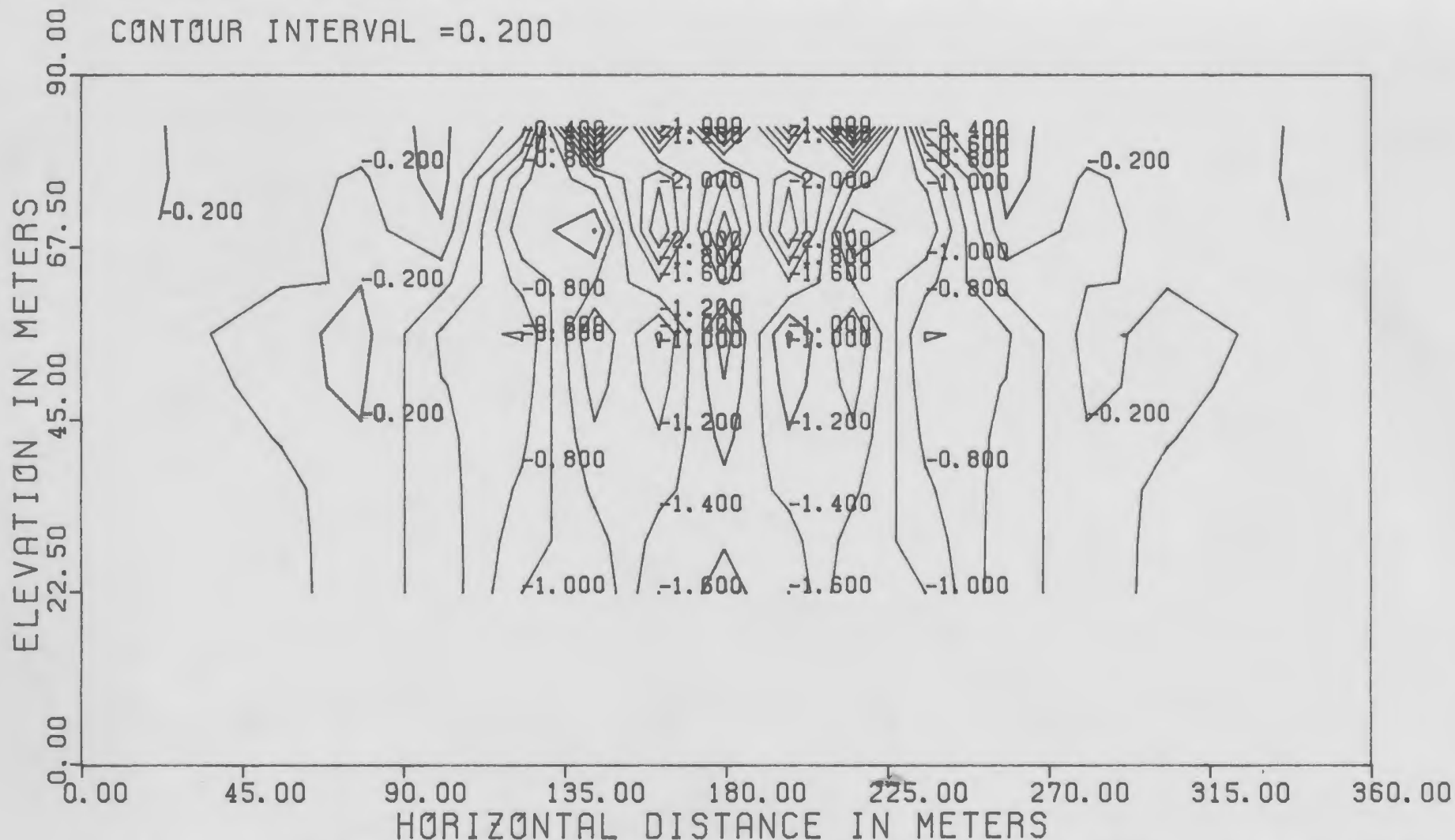


FIG. 6.6 (b) STRESS CONTOURS PLOT FOR PORE PRESSURE
(WITH CAISSON; SUBMERGED WEIGHT ONLY)

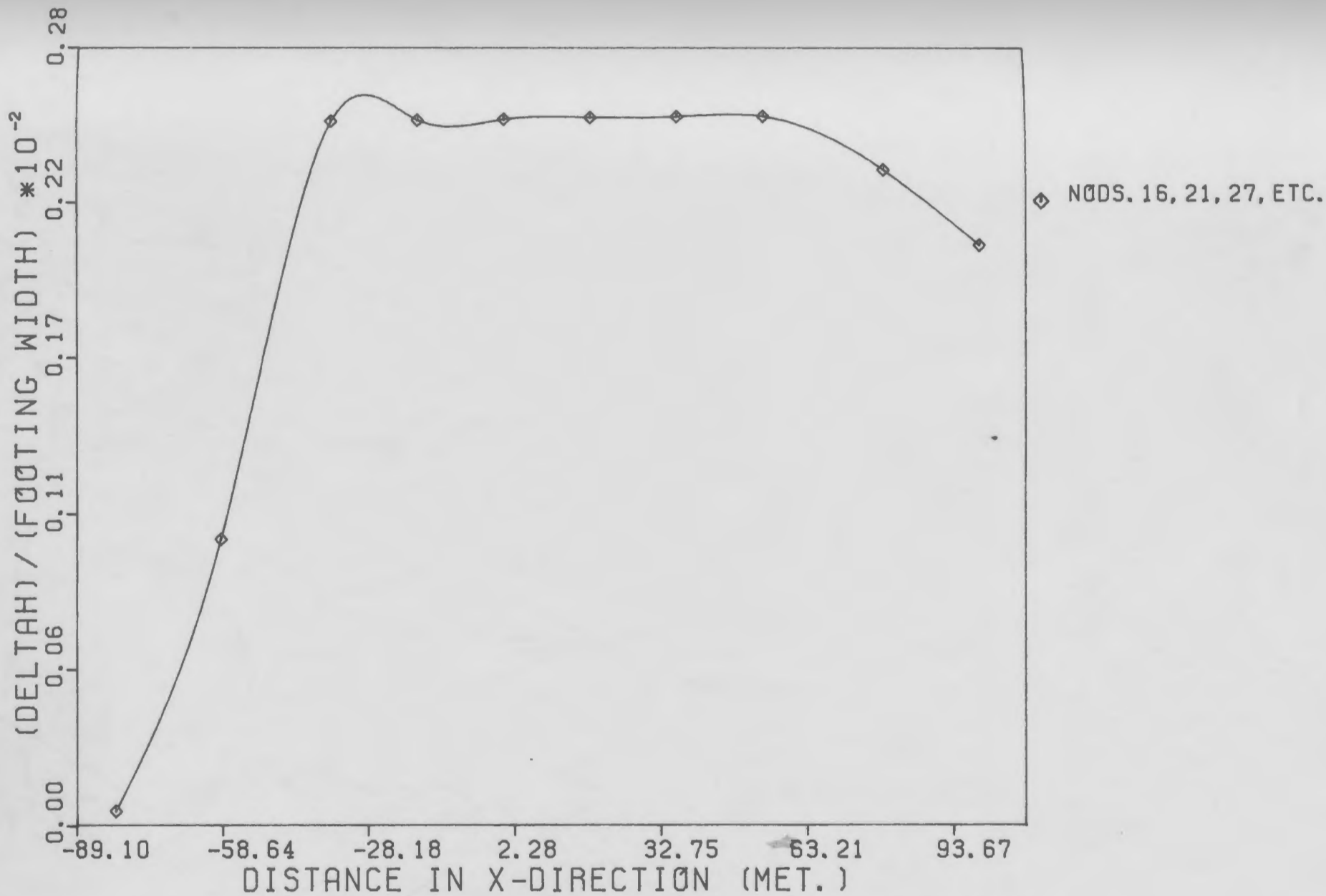
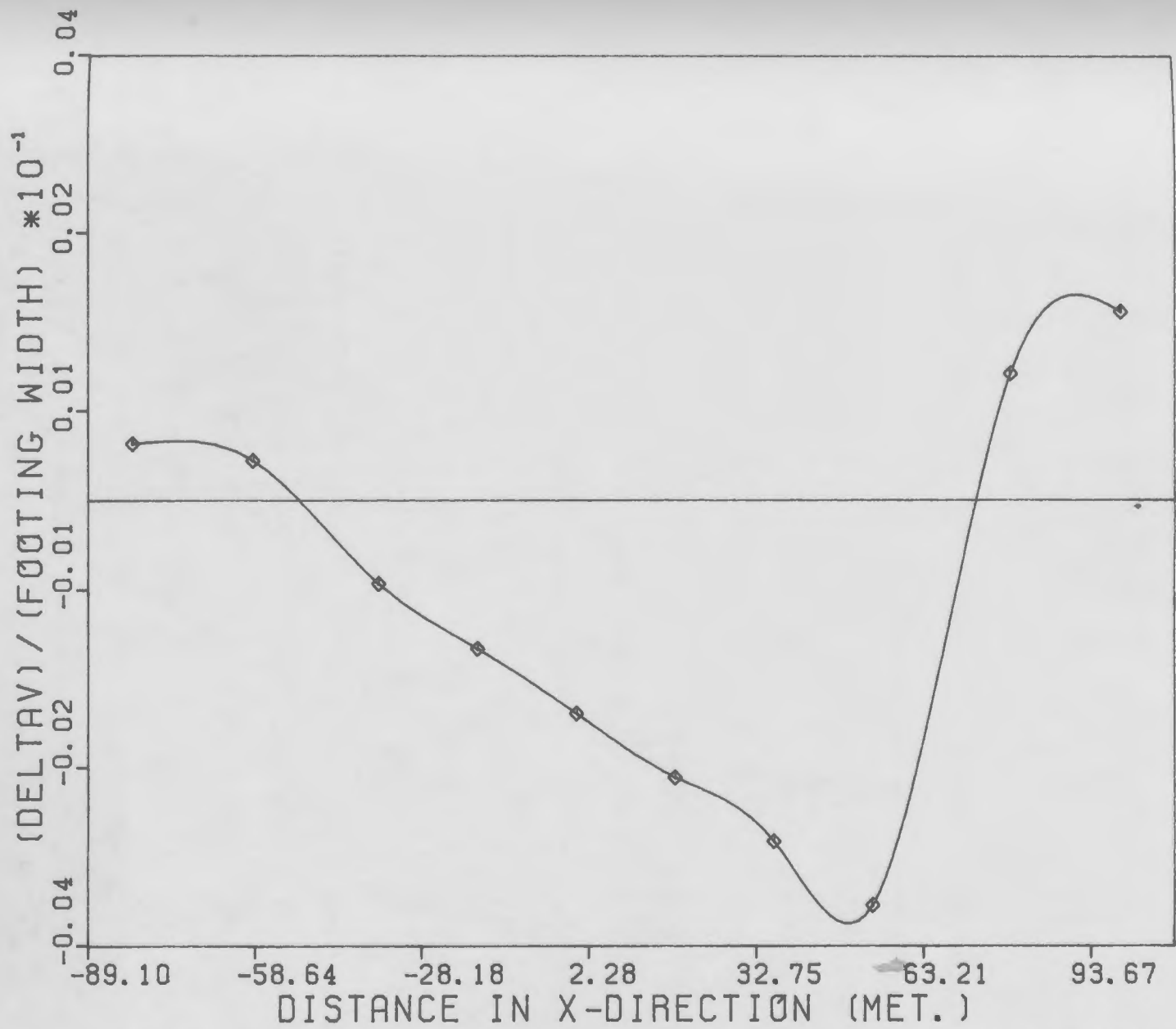


FIG. 6.7 (a) DISTRIBUTION OF HOR. DISPLACEMENT
(WITH CRISSON; INCLINED LOADING)



◇ NODES. 16, 21, 27, ETC.

FIG. 6.7 (b) DISTRIBUTION OF VER. DISPLACEMENT (WITH CAISSON; INCLINED LOADING)

footing. The vertical deformations, δ_V , at nodes 39 and 57 below the caisson are shown in Table 6.3 for zero cycle i.e. due to the submerged weight of the structure. However, the vertical deformation, δ_V , at node 57 becomes almost twice that of node 39 for 1/4 cycle. This is due to the fact that at 1/4 cycle, the peak horizontal force, F_H , acts at the top nodes of the caisson structure causing a moment at the base level. This moment induces additional vertical compressive stresses at elements below the right edge of the caisson which result in substantial increase in the vertical deformation. The horizontal deformations at nodes e.g. 27, 33, 39, 45, 51 and 57 beneath the caisson are 0.25% of the footing width. However, the deformations at nodes, 62 and 67, are 11% and 27% less as compared to the deformation at node, 39. The distribution of shear stress beneath the foundation is shown in Fig. 6.8. It is observed that shear stress intensity is high in elements near the right edge of the footing. For example, the shear stress at element, 46 is 2.25 times higher than that compared to element, 32. This is because of the additional shear stress which is induced near the edge due to the static moment, F_M , besides the effect of the horizontal force, F_H . The comparison of normal stress, σ_y ,

TABLE 6.3 - COMPUTED ELASTIC DEFORMATIONS
AT VARIOUS LOAD CYCLES

Nodes	Loading Cycles	Deformation in meters	
		Horizontal, δ_H	Vertical, δ_V
39	0	-	-0.1580
	1/4	0.2262	-0.1780
57	0	-	-0.1580
	1/4	0.2271	-0.3295

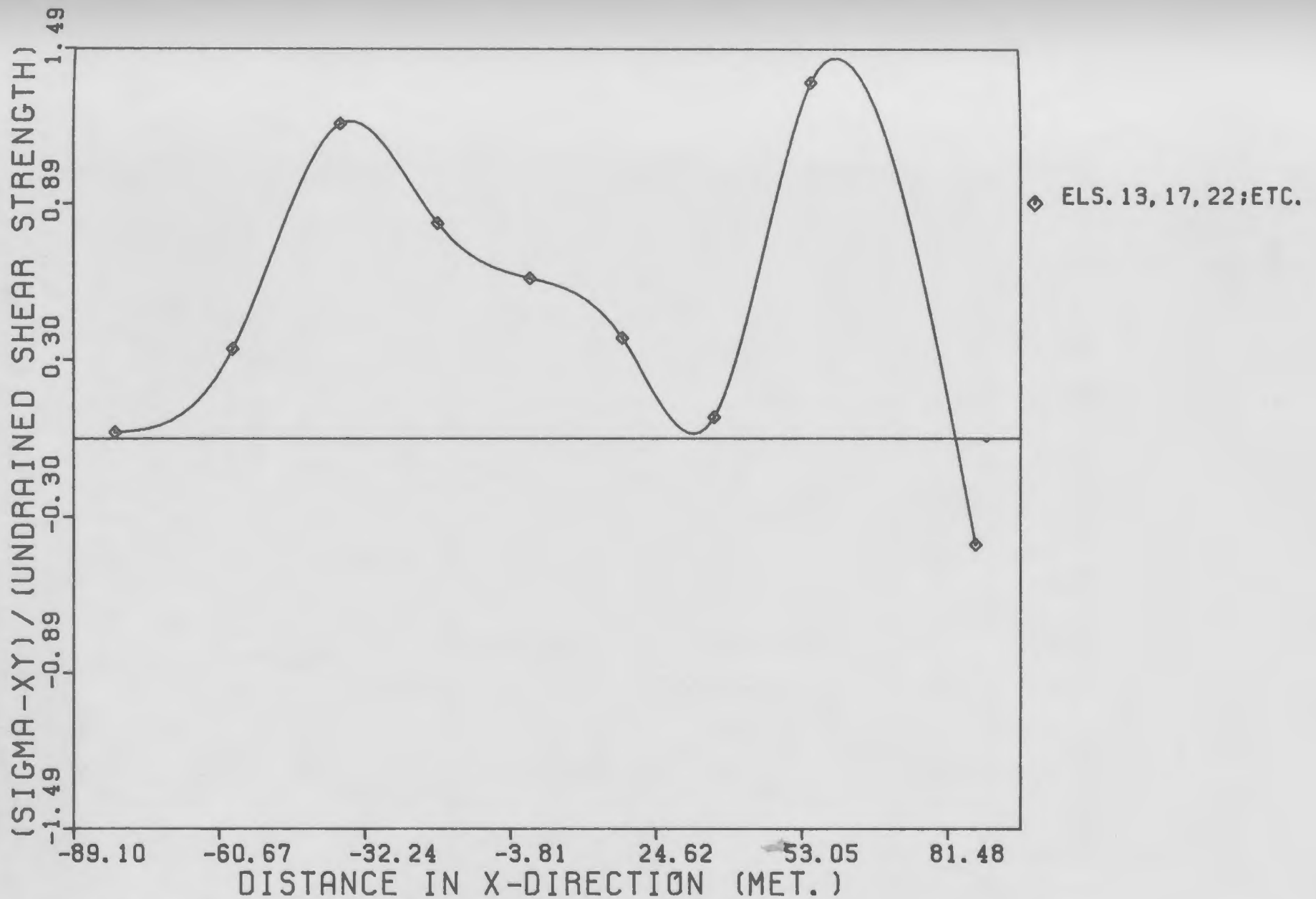


FIG. 6.8 DISTRIBUTION OF SHEAR STRESS
(WITH CRISSON; INCLINED LOADING)

is also made along the depth for the element groups lying on the center line and left of the right edge of the footing. It is seen from Fig. 6.9(a) that the normal stress is higher in elements near the right edge of the caisson. Fig. 6.9(b) depicts the comparison of the distributions of shear stress for the same groups of elements. The contours for shear stress component τ_{xy} are shown in Fig. 6.10 for combined vertical and peak horizontal force which act at the top of the caisson.

6.4.6 Elastic-Plastic Analysis

The nonlinear stress analysis is carried out based on the assumption that the soil behaves as an elastic-perfectly plastic material obeying the Von Mises criteria. The static shakedown load factor obtained in Section 6.4.4 provides the loading domain at shakedown. This load is then applied uniformly at the top nodes of the caisson. The static vertical load, F_V is also applied in a distributed manner at those nodes.

The entire analysis is carried out for 5 cycles of loading i.e. for 60 seconds; 420 load steps are used in order to trace the complete loading path. The uniaxial compressive strength, $\bar{\sigma}$, of soil

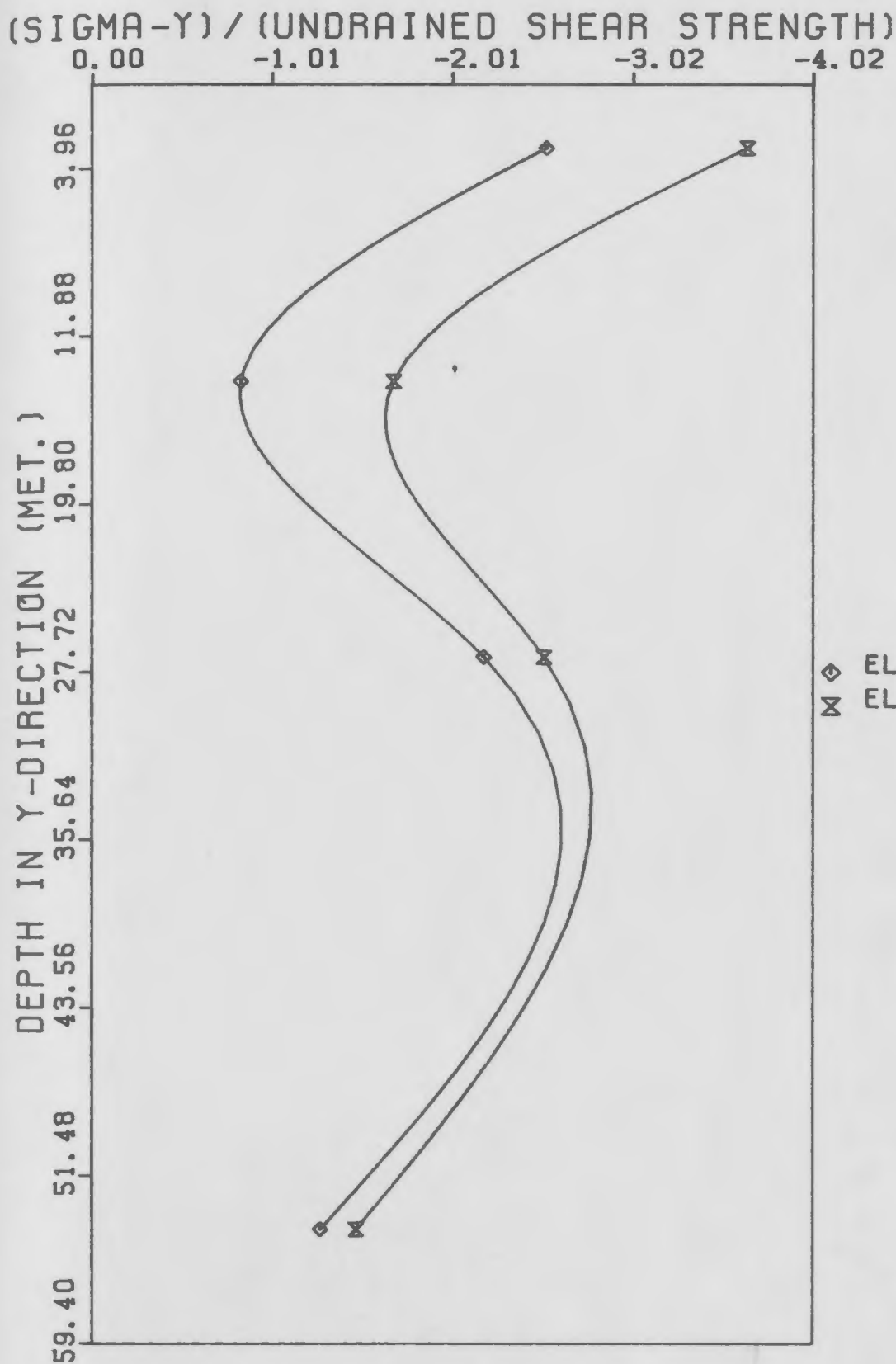


FIG. 6.9 (a) DISTRIBUTION OF NORMAL STRESS
(WITH CAISSON; INCLINED LOADING)

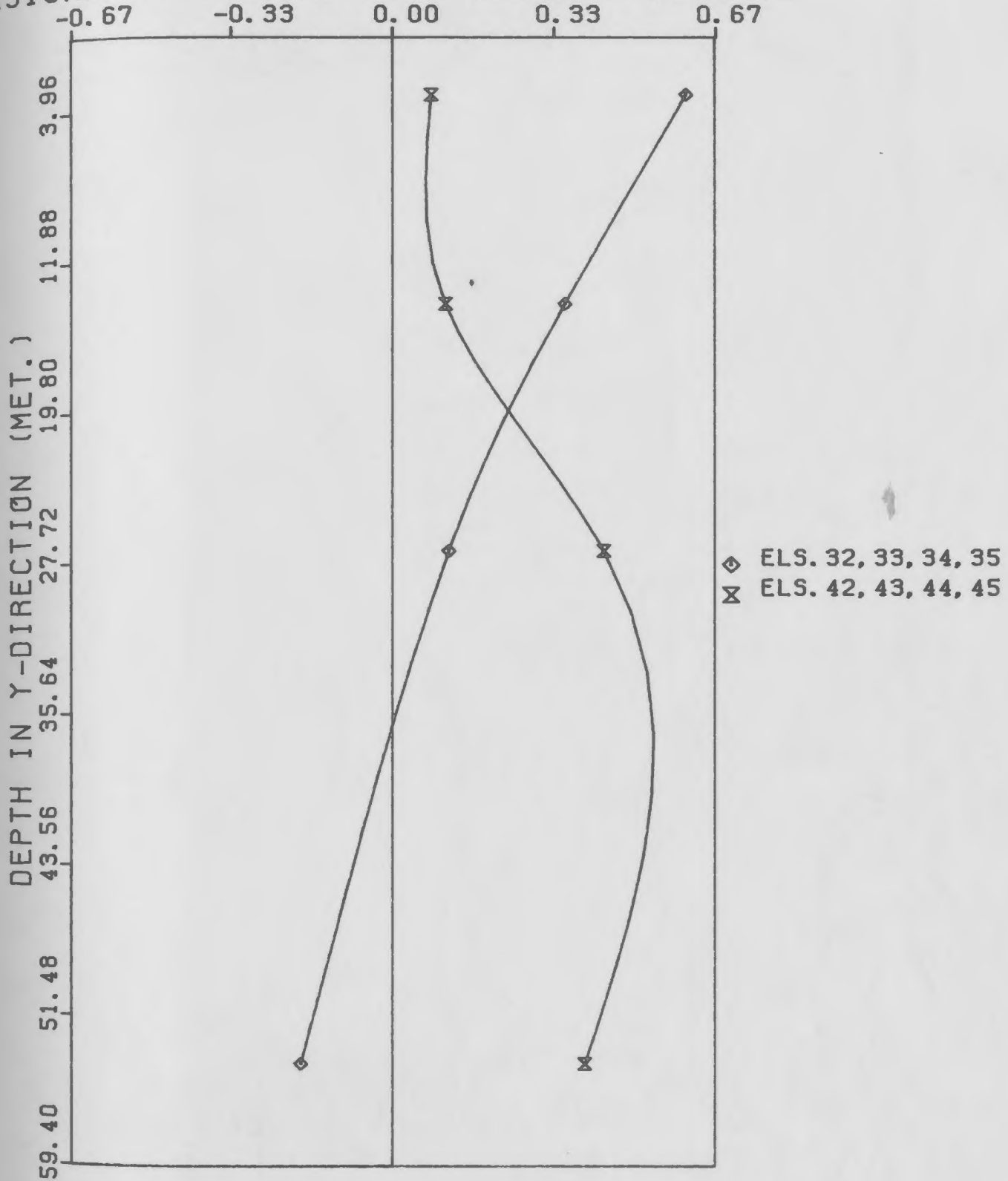
$(\text{SIGMA-XY}) / (\text{UNDRAINED SHEAR STRENGTH})$


FIG. 6.9 (b) DISTRIBUTION OF SHEAR STRESS
(WITH CAISSON; INCLINED LOADING)

(SIGMA-XY) / (UNDRAINED SHEAR STRENGTH)

CONTOUR INTERVAL = 0.200

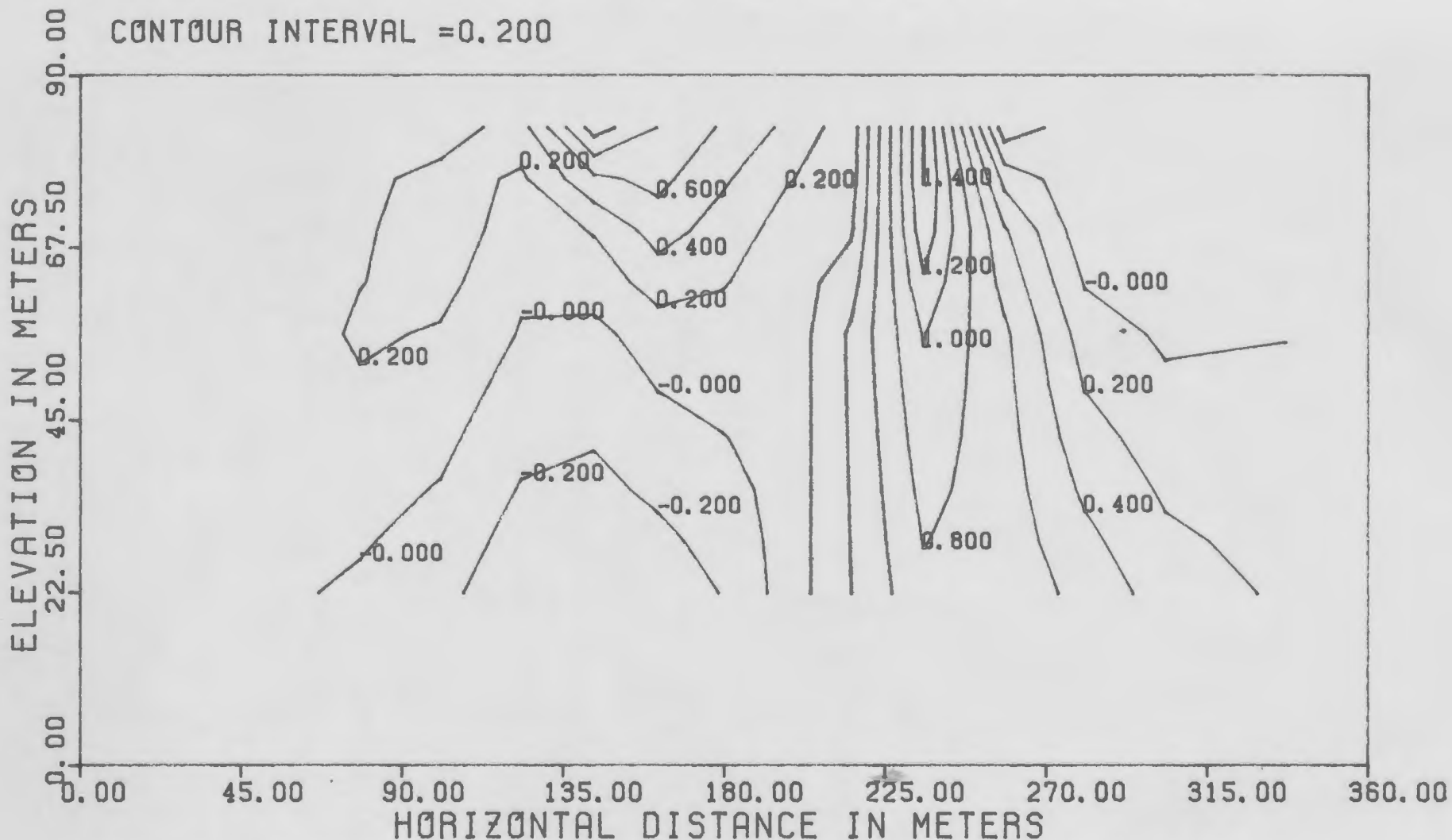


FIG. 6.10 STRESS CONTOURS PLOT FOR (SIGMA-XY)
(WITH CAISSON; INCLINED LOADING)

beneath the foundation is related to the undrained unsoftened static shear strength, s_u , as $\sqrt{3} s_u$ for plane strain condition. The caisson is modelled with the compressive strength of concrete and a low value of angle of internal friction, ϕ_u . The properties of the soil and the concrete are shown in Table 6.4.

The horizontal and vertical deformations beneath the caisson due to the cyclic wave loading is shown in Figs. 6.11 and 6.12 for various loading cycles. They show clearly that the deformations along horizontal and vertical directions have stabilized (shakedown) after second loading cycle. Virtually there is no increase in the deformations in the subsequent cycles. This is in agreement qualitatively with the experimental observation made by Rowe (1975) on small scale model tests and described in Section 2.4.2.

Table 6.5 shows the horizontal and vertical deformations for two nodes, 39 and 57, at cycles 0, 1/4, 1, 3 1/4, 4, 4 1/4 and 5 cycles. The vertical deformation, δ_v , due to the submerged weight of the structure is 0.1580 meter ie. 0.18% of the width of the footing. It is seen from the above Table 6.5 that vertical deformations at nodes 39

TABLE 6.4 - MATERIAL PROPERTIES FOR THE
CAISSON STRUCTURE AND THE
SATURATED SOIL MEDIUM

Notations	Caisson	Soil
λ	4200.0 MPa	14.7 MPa
μ	9800.0 MPa	9.80 MPa
α	-	1.0
K_w	-	2000.0 Mpa
ρ	22.0 KN/m ³	19.0 KN/m ³
ρ_f	-	9.8 KN/m ³
n	-	0.30
v_d	0.15	0.30
v_u	-	0.499
c'	15.00 MPa	100.0 KPa
ϕ_u	10 ⁰	-

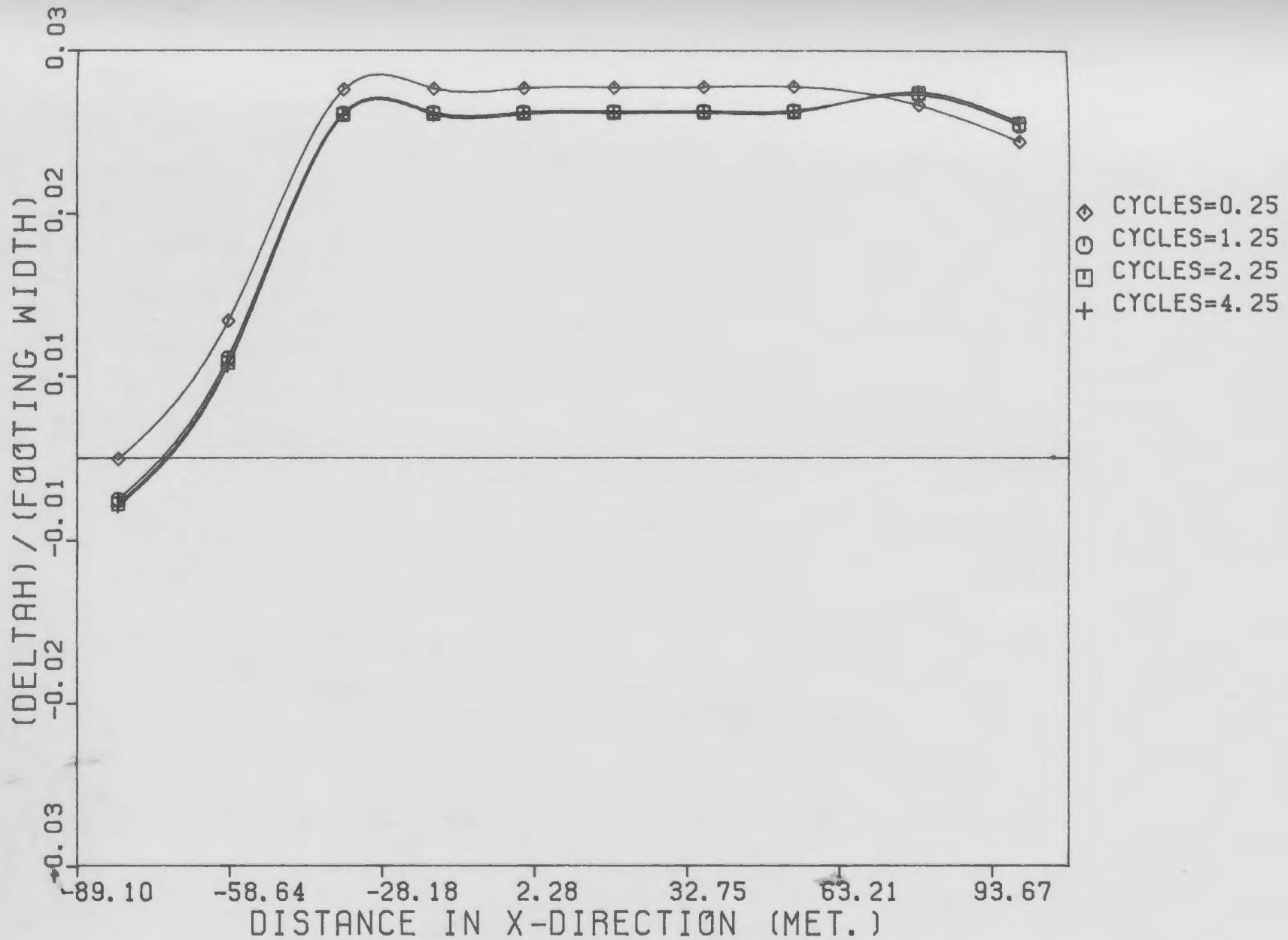


FIG. 6.11 DISTRIBUTION OF HOR. DISPLACEMENT
(STIFF FOOTING; ELASTIC PLASTIC ANALYSIS)

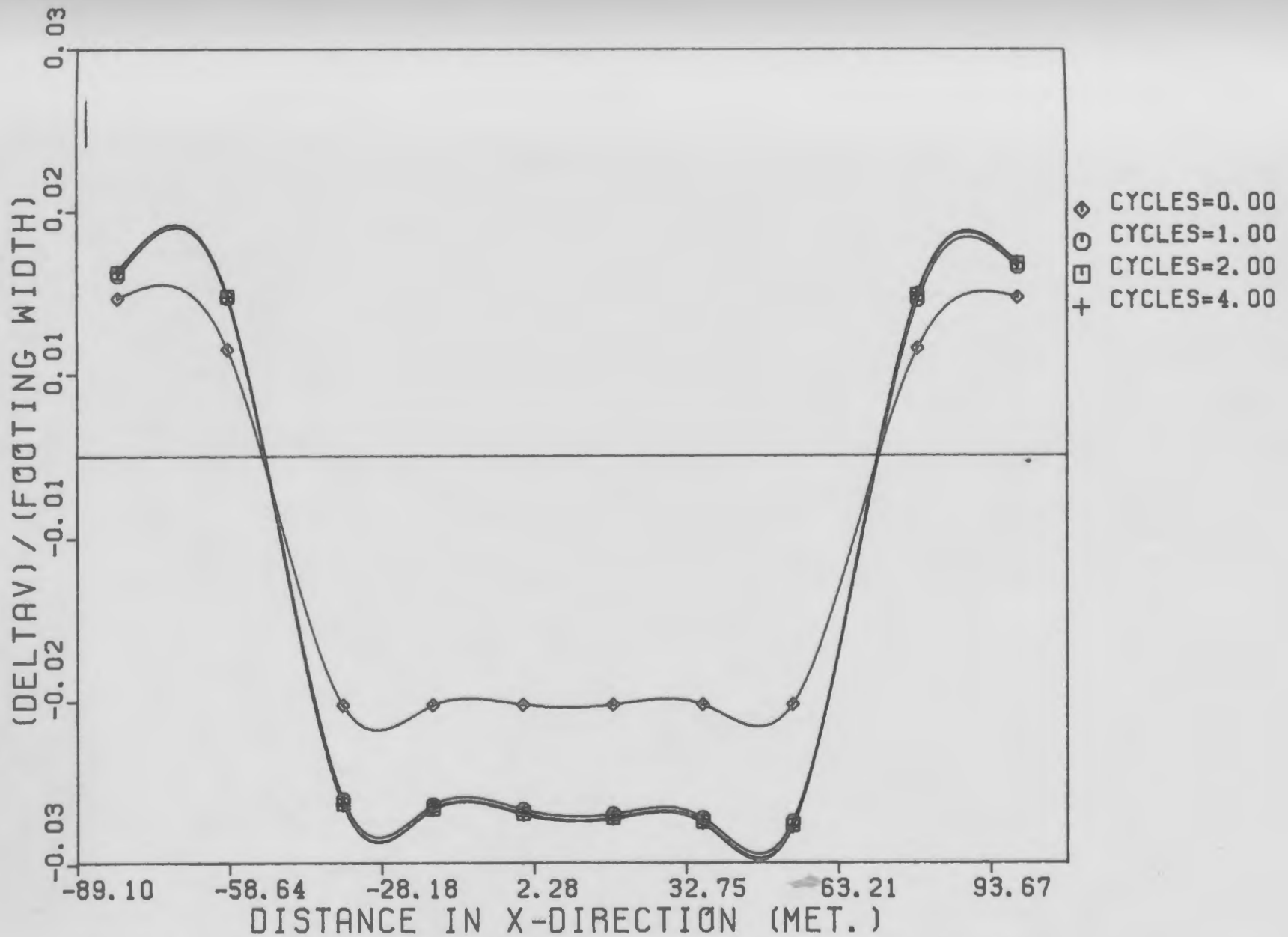


FIG. 6.12 DISTRIBUTION OF VER. DISPLACEMENT
 (STIFF FOOTING; ELASTIC PLASTIC ANALYSIS;)

TABLE 6.5 - COMPUTED DEFORMATIONS (IN METERS) FOR
LINEAR AND NONLINEAR ANALYSES

	Types of Analyses	N = 0		N = 1/4		N = 1		N = 3 1/4		N = 4		N = 4 1/4		N = 5	
		δ_H	δ_V	δ_H	δ_V	δ_H	δ_V	δ_H	δ_V	δ_H	δ_V	δ_H	δ_V	δ_H	δ_V
39	Elastic	-	-0.1580	.2262	-0.1780	-	-	-	-	-	-	-	-	-	-
	Elastic-Plastic	-	-0.1580	.2500	-0.1850	-	-0.2300	.2324	-0.2000	-	-0.2280	.2322	-.2021	-	-.2300
57	Elastic	-	-0.1580	.2271	-.3295	-	-	-	-	-	-	-	-	-	-
	Elastic-Plastic	-	-0.1580	.2500	-.3800	-	-0.2327	.2332	-.3660	-	-0.2361	0.2330	-.3663	-	-.2360

and 57 have stabilized to a value of 0.2291 meter and 0.2314 meter respectively. These deformations are 45% higher than the initial settlement at zero load cycle.

The soil elements 22, 27 and 46 beneath the caisson yielded at peak horizontal load. The distribution of shear stress beneath the caisson for selected elements is shown in Fig. 6.13. Fig. 6.14 depicts the distribution of shear stress along the depth for the elements lying just on the centerline. Fig. 6.15 shows similar distributions along the depth for those elements near the left of the right edge of the caisson structure. It is seen that shakedown has taken place with a redistribution of shear stress once the elements have yielded and as the load passes from first cycle to the subsequent cycle.

6.4.7 Comparison of Elastic and Elastic-Plastic Analyses

The comparison of the two static analyses, viz elastic and elastic-plastic are presented at 1/4 load cycle when the horizontal force is at its peak value. Figs. 6.16(a) and (b) show the comparisons for horizontal and vertical deformations for the two analyses. For node 57, the horizontal

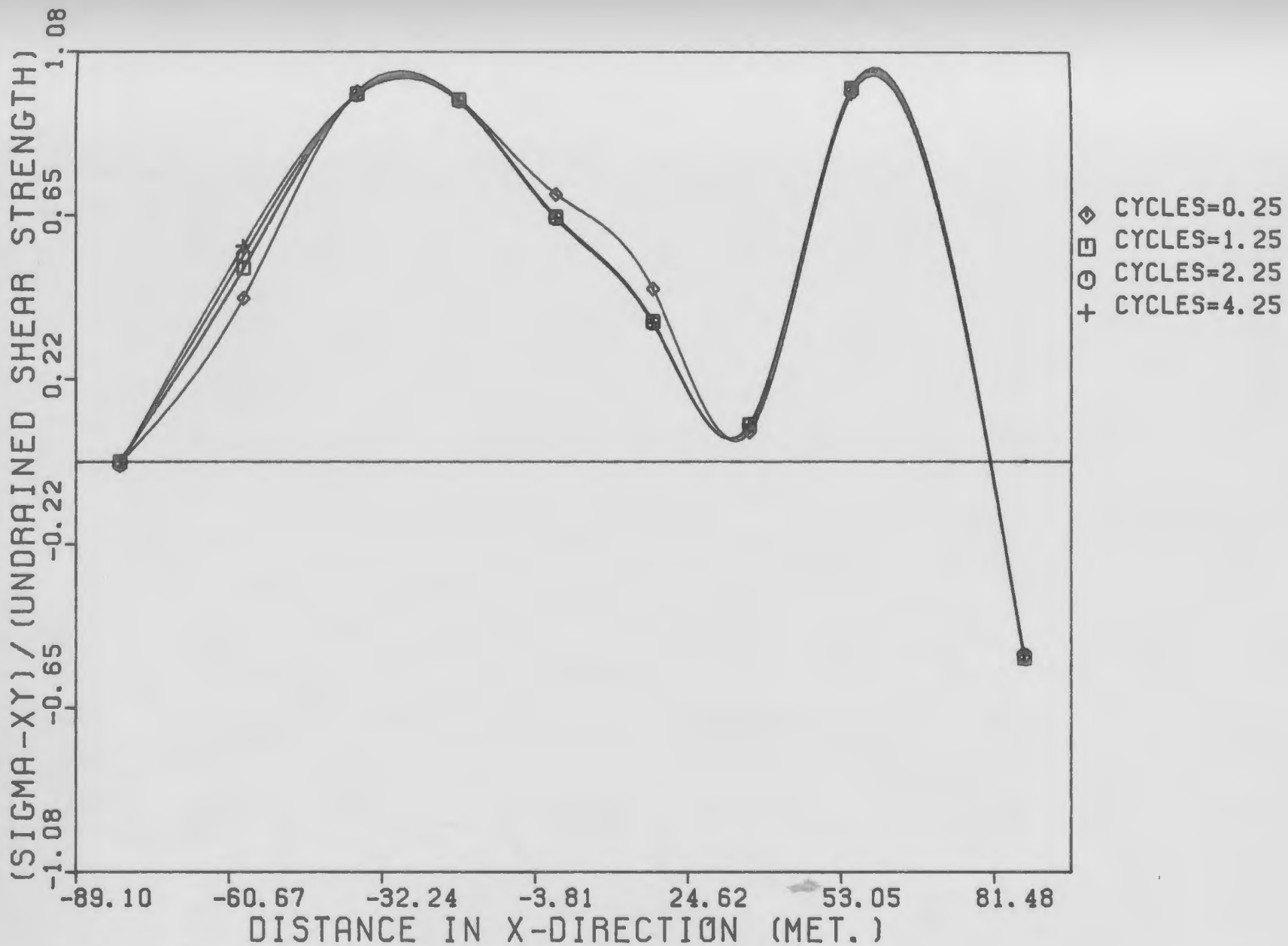


FIG. 6.13 DISTRIBUTION OF SHEAR STRESS
 (WITH CAISSON; ELASTIC PLASTIC ANALYSIS)

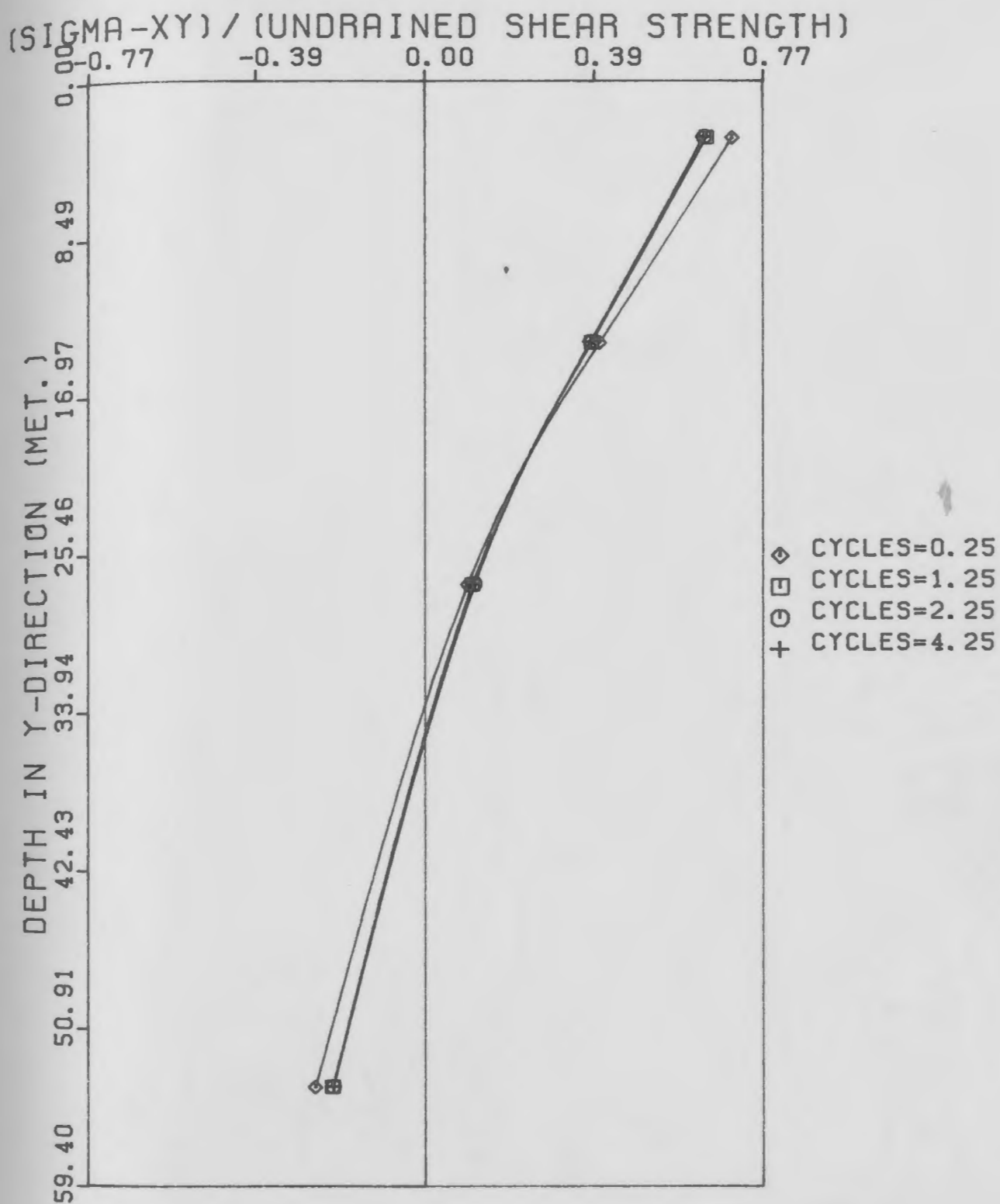


FIG. 6.14 DISTRIBUTION OF SHEAR STRESS
 (WITH CRISSON; NONLINEAR ANALYSIS)
 (ELEMENTS 32, 33, 34, 35)

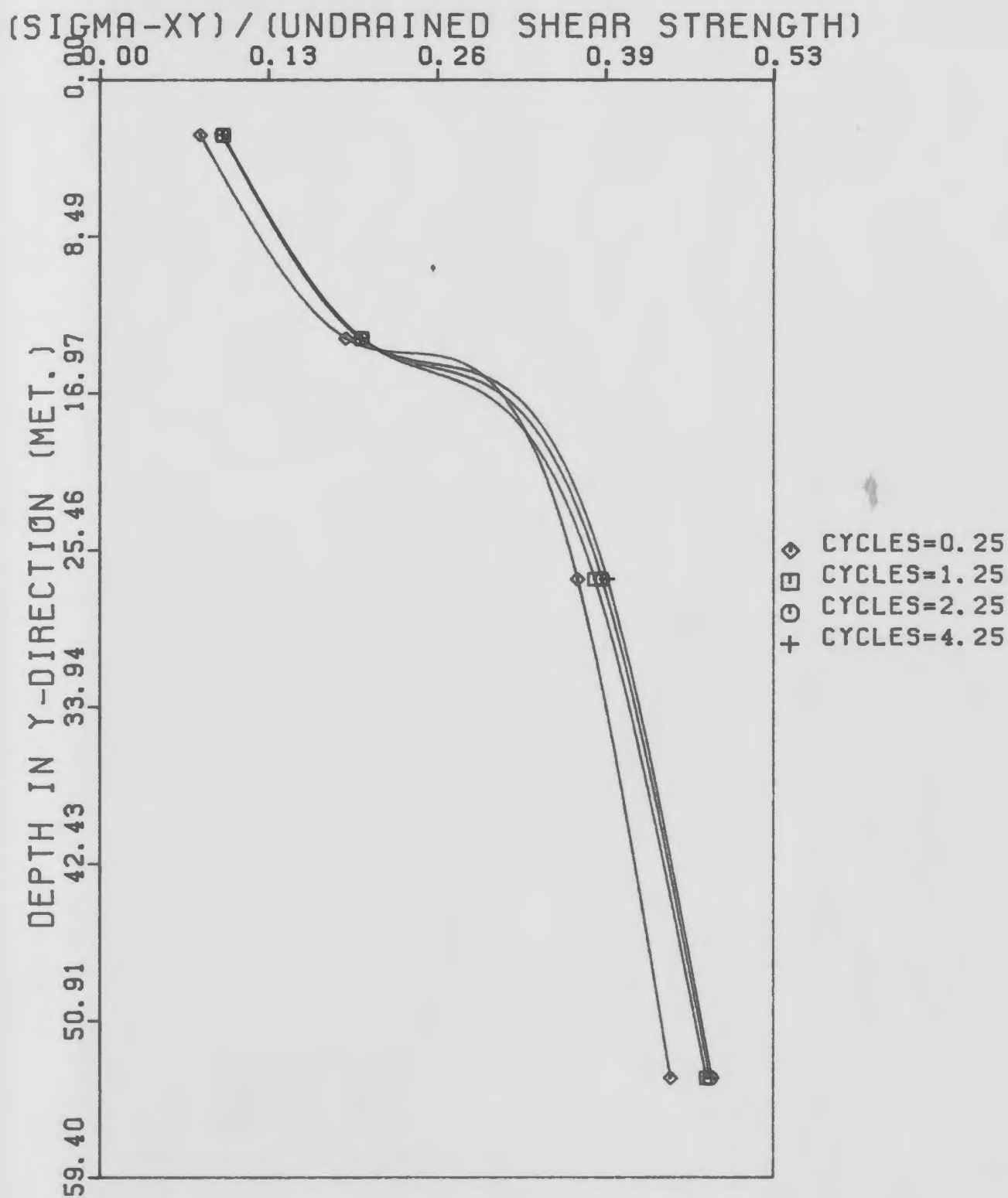


FIG. 6.15 DISTRIBUTION OF SHEAR STRESS
 (WITH CRISSON; NONLINEAR ANALYSIS)
 (ELEMENTS 42, 43, 44, 45)

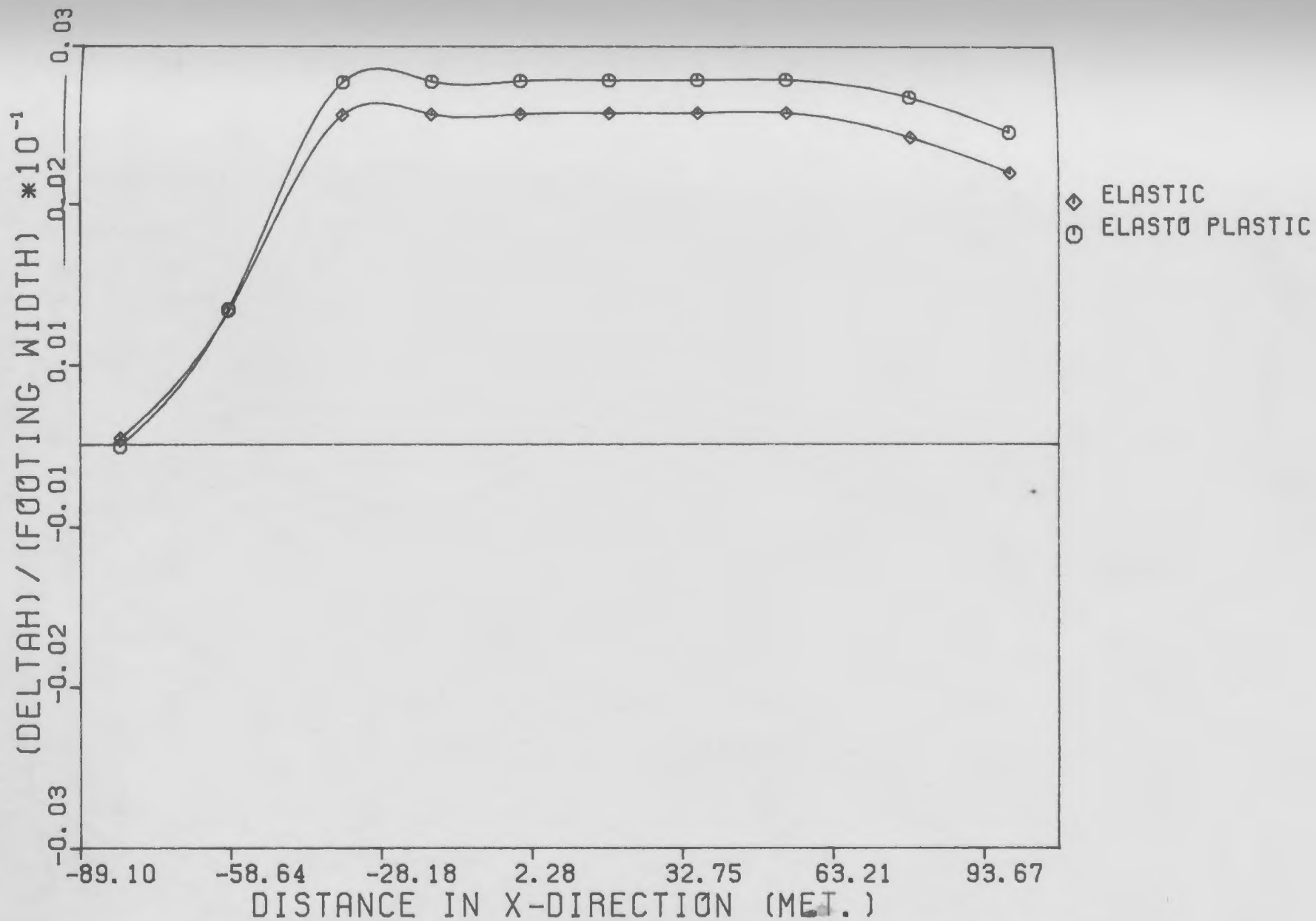


FIG. 6.16 (a) DISTRIBUTION OF HOR. DISPLACEMENT
(WITH CAISSON; COMPARISON OF ANALYSES)
(AT 1/4 CYCLE)
(NODES 16, 21, 27, 33, 39, 45, 51, 57, 62, 67 ETC)

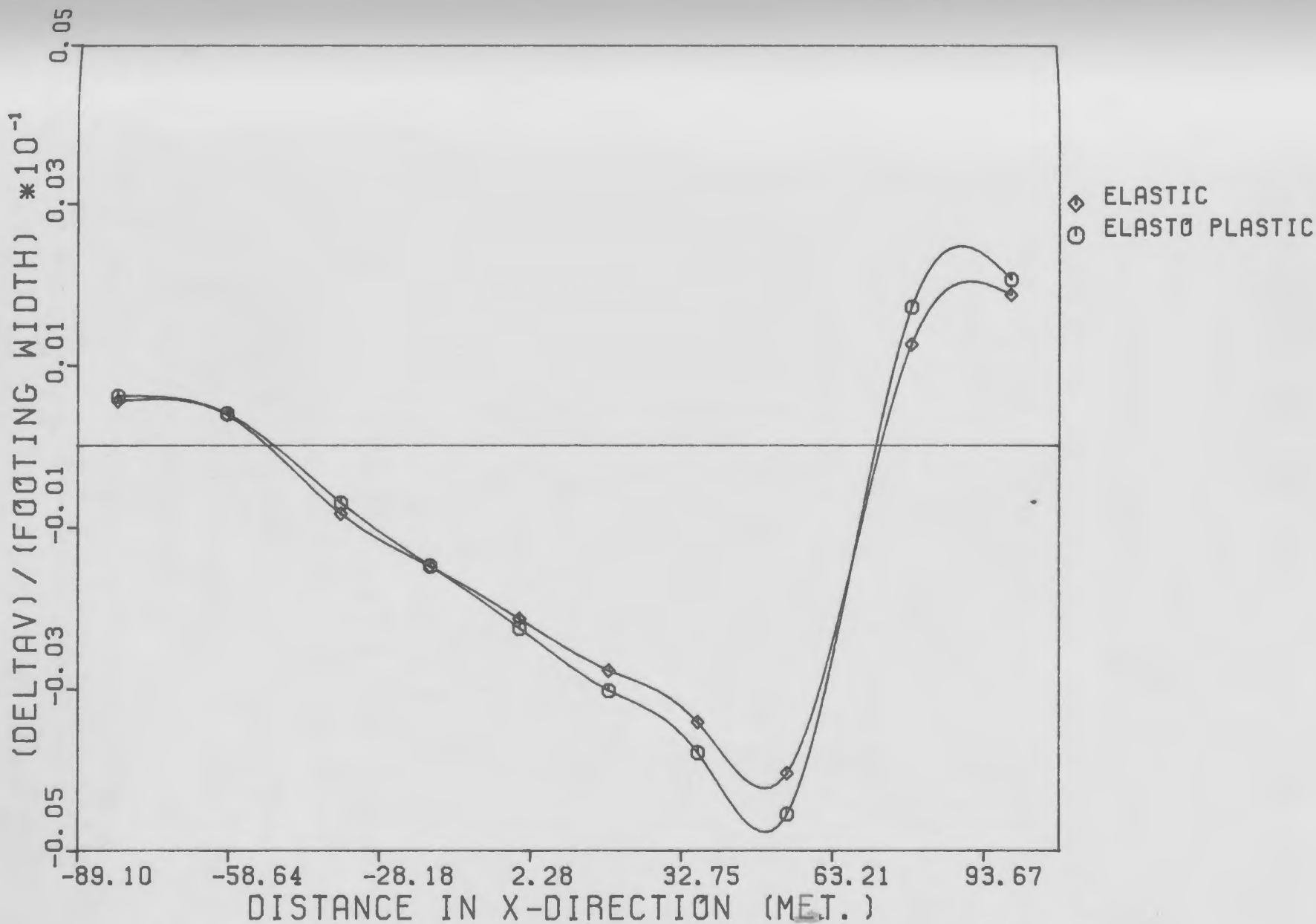


FIG. 6.16 (b) DISTRIBUTION OF VER. DISPLACEMENT
 (WITH CAISSON; COMPARISON OF ANALYSES)
 (AT 1/4 CYCLE)
 (NODES 16, 21, 27, 33, 39, 45, 51, 57, 62, 67 ETC)

and vertical deformations from elastic-plastic analysis are 12% and 13% higher than those obtained by elastic analysis, since some soil elements beneath the caisson have yielded. Fig. 6.17 shows the distribution of δ_H along the depth. Fig. 6.18 compares the shear stress distribution beneath the foundation for the two analyses. The shear stress predicted by elastic-plastic analysis is lower than that predicted by the elastic analysis. For example, the elastic shear stress at element, 46 is 45% higher than obtained from a nonlinear analysis. Fig. 6.19 compares the shear stress along the depths for elements lying along the centerline.

6.5 Dynamic Analysis of a Gravity Type Offshore Foundation

The present investigation considers the dynamic interaction between the caisson structure of a gravity base offshore platform and its foundation soil which consists of saturated clay underlaid by a rigid base. The caisson structure is assumed to be linear whereas the nonlinear behaviour of the soil is taken into account by two different types of analyses: a) equivalent linear model and b) elastic-perfectly plastic model. The cyclic wave

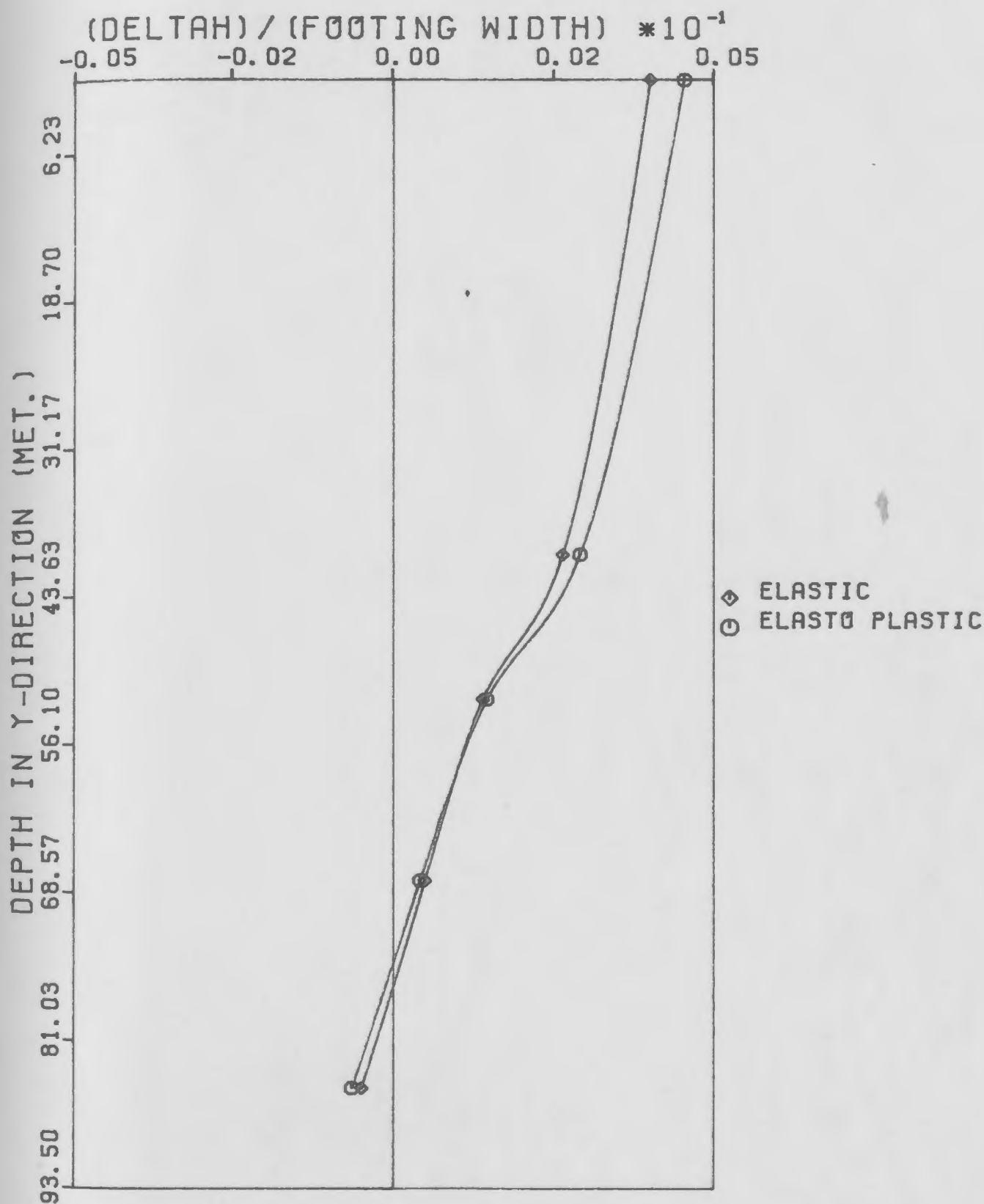


FIG. 6. 17 DISTRIBUTION OF HOR. DISPLACEMENT
(WITH CAISSON; COMPARISON OF ANALYSES)
(NODES 38, 39, 40, 41, AND 42)

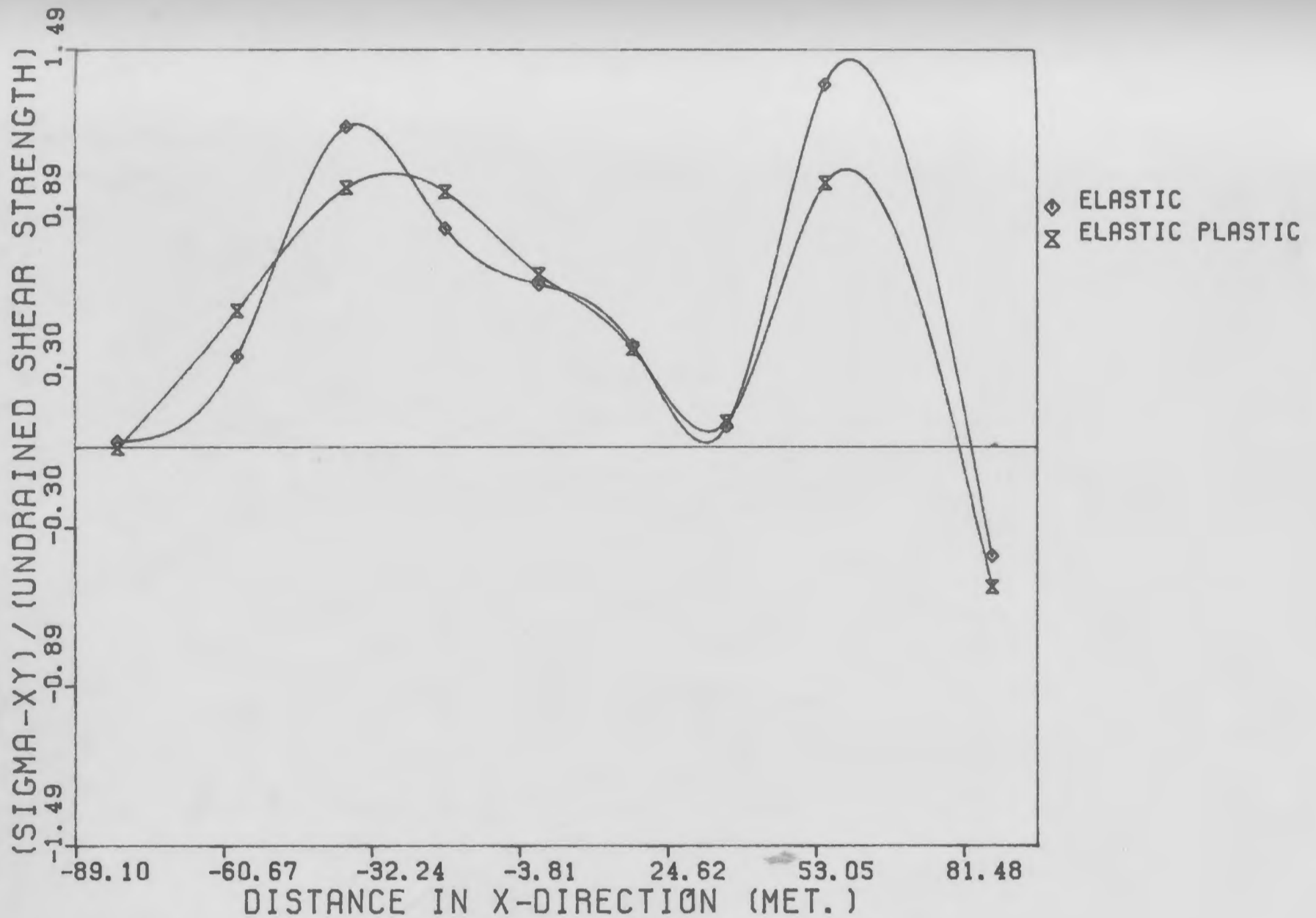


FIG. 6.18 DISTRIBUTION OF SHEAR STRESS
(WITH CRISSON; COMPARISON OF ANALYSES)
(AT 1/4 CYCLE)
(ELEMENTS 13, 17, 22, 27, 32, 37, 42, 46, 50)

(SIGMA-XY) / (UNDRAINED SHEAR STRENGTH)

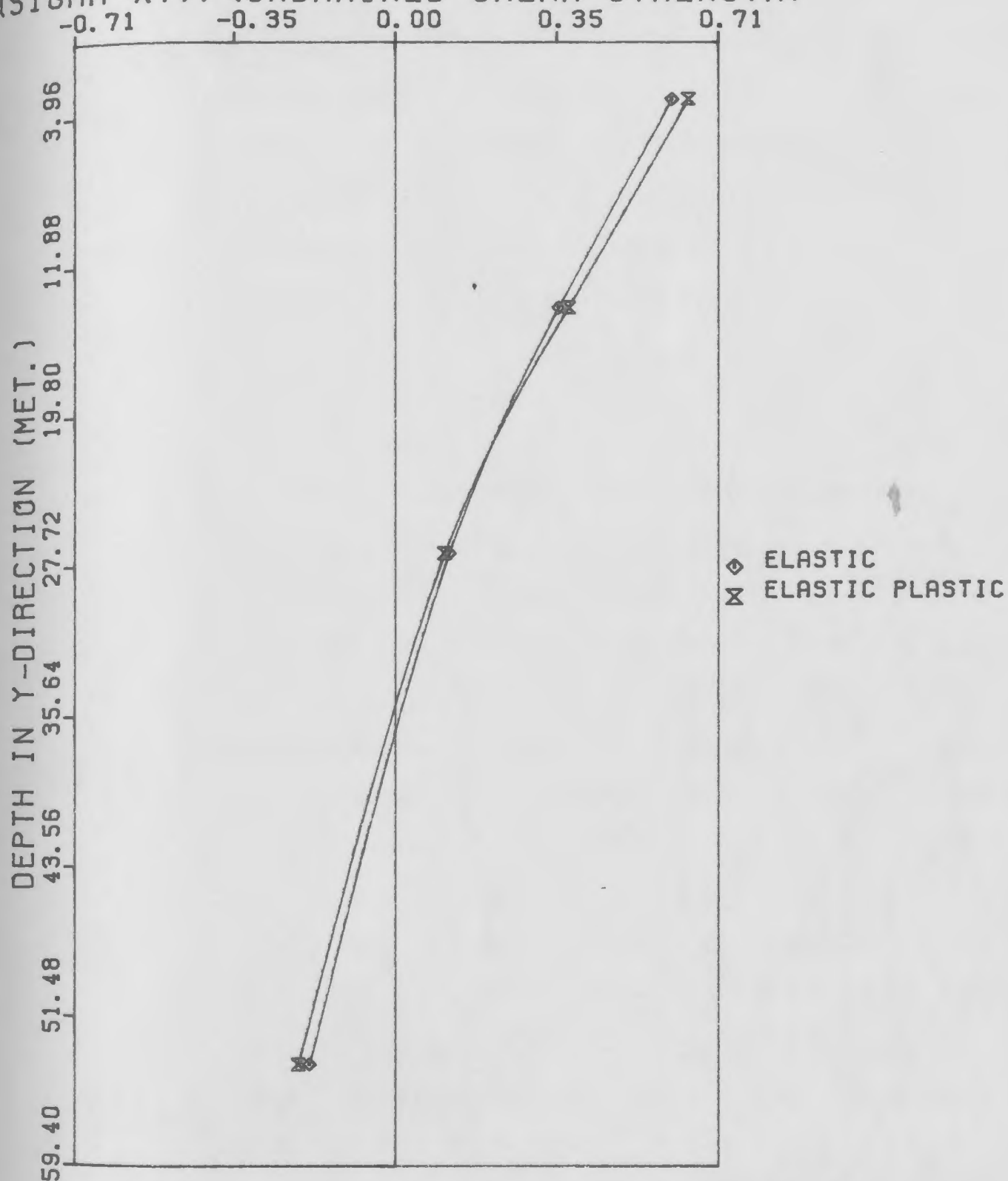


FIG. 6.19 DISTRIBUTION OF SHEAR STRESS
 (WITH CRISSON; COMPARISON OF ANALYSES)
 (32, 33, 34, 35; CYCLE=0.25)

loading function is simulated here as sinusoidal forcing function, with a 12 seconds period, acting at the top of the caisson's nodes thus causing a dynamic cyclic moment at the foundation level. The submerged weight, F_V , is assumed to be static and remains constant during the entire loading cycle. ' .

6.5.1 Dynamic Shakedown Analysis

In order to carry out the dynamic shakedown analysis, the elastic stress responses of the structure-foundation system have to be determined. The time history of response quantities are computed by an implicit step-by-step integration scheme which is primarily dependent on the time step interval (Δt) chosen. The time step interval is a function of the system frequencies and Table 6.6 shows the computed frequencies of the caisson-foundation system for first ten modes under drained and undrained conditions. The drained condition is simulated by setting the bulk modulus of water as zero. The fundamental frequency in the drained state is only 12% less than that in undrained state. However, for higher modes, the difference is considerably large. For example, the frequency at eighth mode for drained state is 31% less than

TABLE 6.6 - UNDRAINED AND DRAINED
FREQUENCIES OF THE CAISSON-
FOUNDATION SYSTEM

MODE NUMBERS	UNDRAINED FREQUENCIES (rad/sec)	DRAINED FREQUENCIES (rad/sec)
1	1.53	1.34
2	2.23	1.86
3	2.24	1.94
4	2.93	2.42
5	2.96	2.44
6	3.20	2.75
7	3.73	2.88
8	3.80	2.90
9	4.33	3.14
10	4.41	3.63

that in undrained state. This is because of the fact that under undrained condition, incompressibility constraint on volumetric mode of deformation introduces additional stiffness.

Although theoretically one should take the period associated with the highest mode, it is known from practice that higher modes contribute very little to the response process. Besides, soil foundation system has a typical frequency ratio

$$\omega^* = \frac{\omega_0}{\omega_1} = 0.34$$

where

$$\omega_0 = \text{wave frequency} = 0.523 \text{ rad/sec.}$$

and

$$\begin{aligned} \omega_1 &= \text{fundamental frequency of the} \\ &\quad \text{structure-foundation system} \\ &= 1.53 \text{ rad/sec. (Refer Table 6.6)} \end{aligned}$$

The time step of 0.1429 second is chosen based on one tenth of the period associated with the tenth mode of the structure-foundation system. This will result in 420 time steps in order to describe the five cycles of loading (60 seconds duration) adequately.

The Rayleigh damping coefficients are computed based on the specified modal damping ratios in

first and second mode. The critical damping values are assumed as 15% and 20% in mode 1 and mode 2 respectively. Based on the above values, the Rayleigh damping coefficients, B_1 and B_2 , for the caisson and the soil medium are computed and shown in Table 6.7.

6.5.1.1 Dynamic Elastic Analysis

The elastic dynamic response is computed first with zero initial conditions for five cycles of loading and the displacements and velocities for each node are saved at the end of the loading process.

Computed displacements and velocities which have been saved are used as fictitious initial conditions, \bar{U}^* , and $\dot{\bar{U}}^*$ and the normalized maximum stress response vector is computed from Eqn. (4.43). Finally, the minimization of Eqn. (4.55) is carried out once again using the above stress vector and the shakedown load factor, β_s , obtained as 1.15. This is 10% lower than the corresponding value obtained from the static analysis (1.28). This is because of the fact that stress amplification has occurred at the frequency ratio of $\omega^* = 0.34$ and as such the shakedown load factor has decreased. Based on this load factor, $\beta_s =$

TABLE 6.7 - RAYLEIGH DAMPING COEFFICIENTS
FOR CAISSON STRUCTURE AND
SOIL MEDIUM

Rayleigh Damping Coefficients	Caisson	Soil
B ₁	-0.340×10^{-2}	0.739×10^{-1}
B ₂	0.276×10^{-1}	0.1650

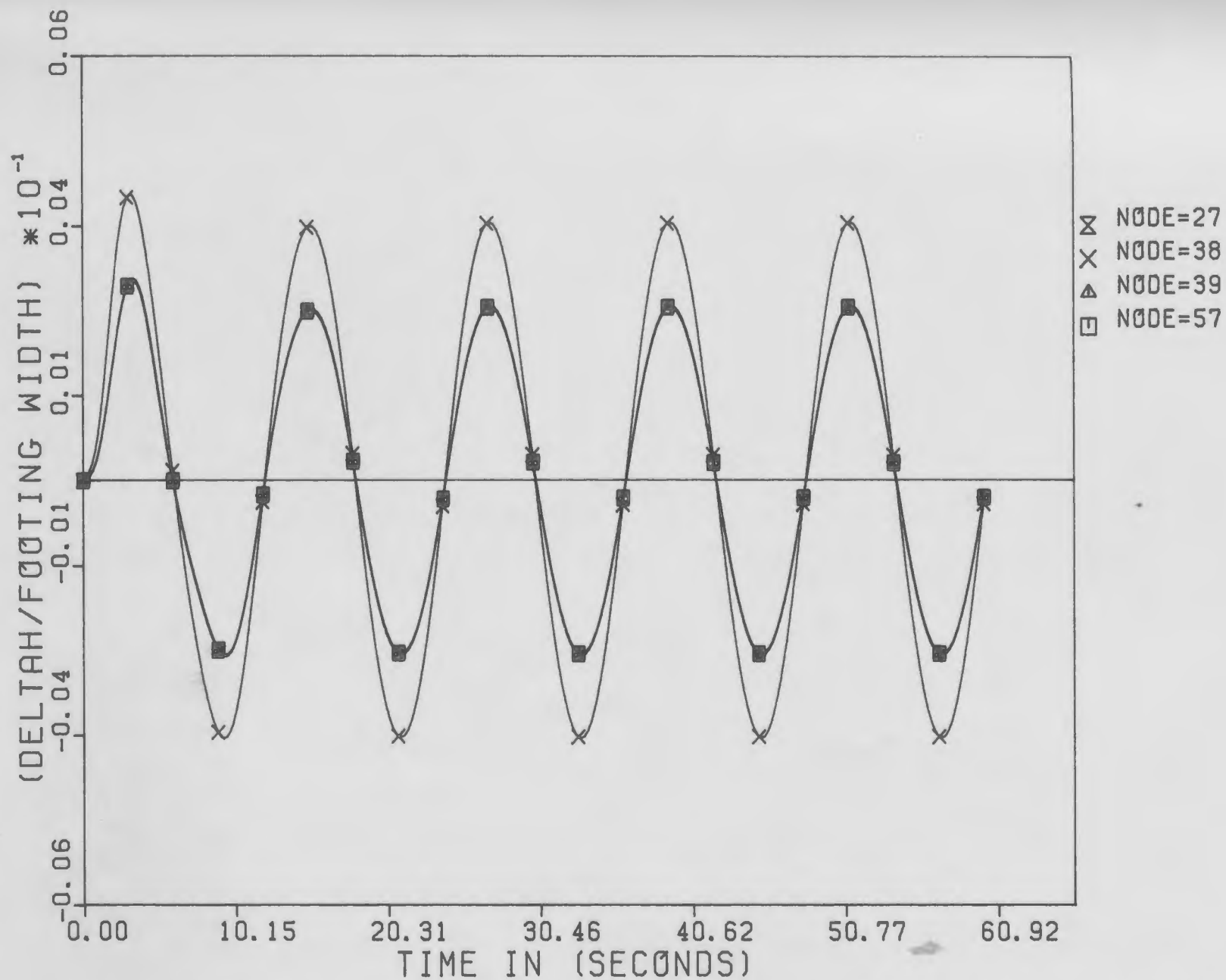


FIG. 6.20 (a) TIME HISTORY PLOT OF HOR. DISPLACEMENT (WITH CAISSON; DYNAMIC ELASTIC ANALYSIS)

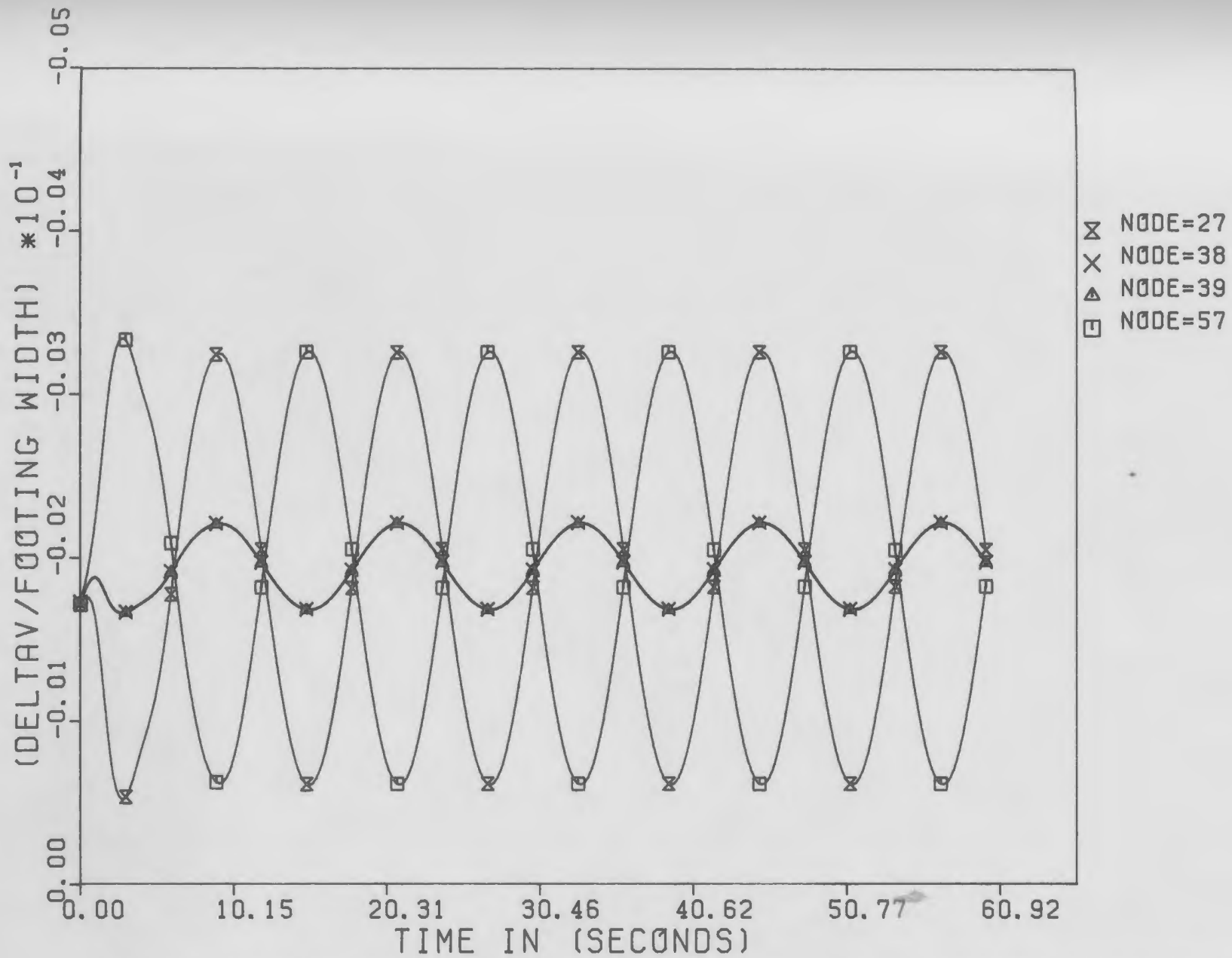


FIG. 6.20 (b) TIME HISTORY PLOT OF VER. DISPLACEMENT
(WITH CAISSON; DYNAMIC ELASTIC ANALYSIS)

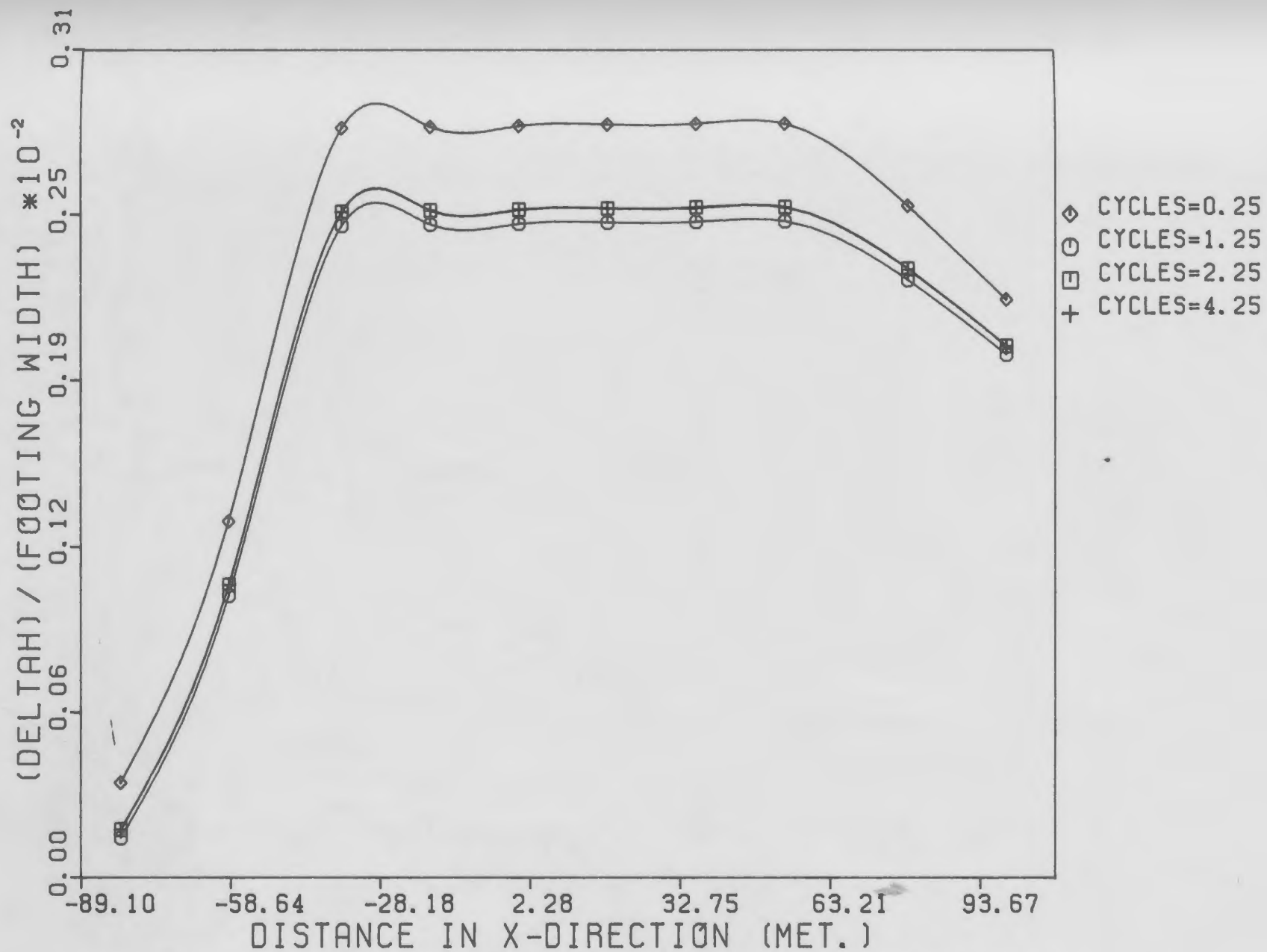


FIG. 6.21 (a) DISTRIBUTION OF HOR. DISPLACEMENT
 (WITH CAISSON; DYNAMIC ELASTIC ANALYSIS)
 (NODES 16, 21, 27, 33, 39, 45, 51, 57, 62, 67 ETC;)

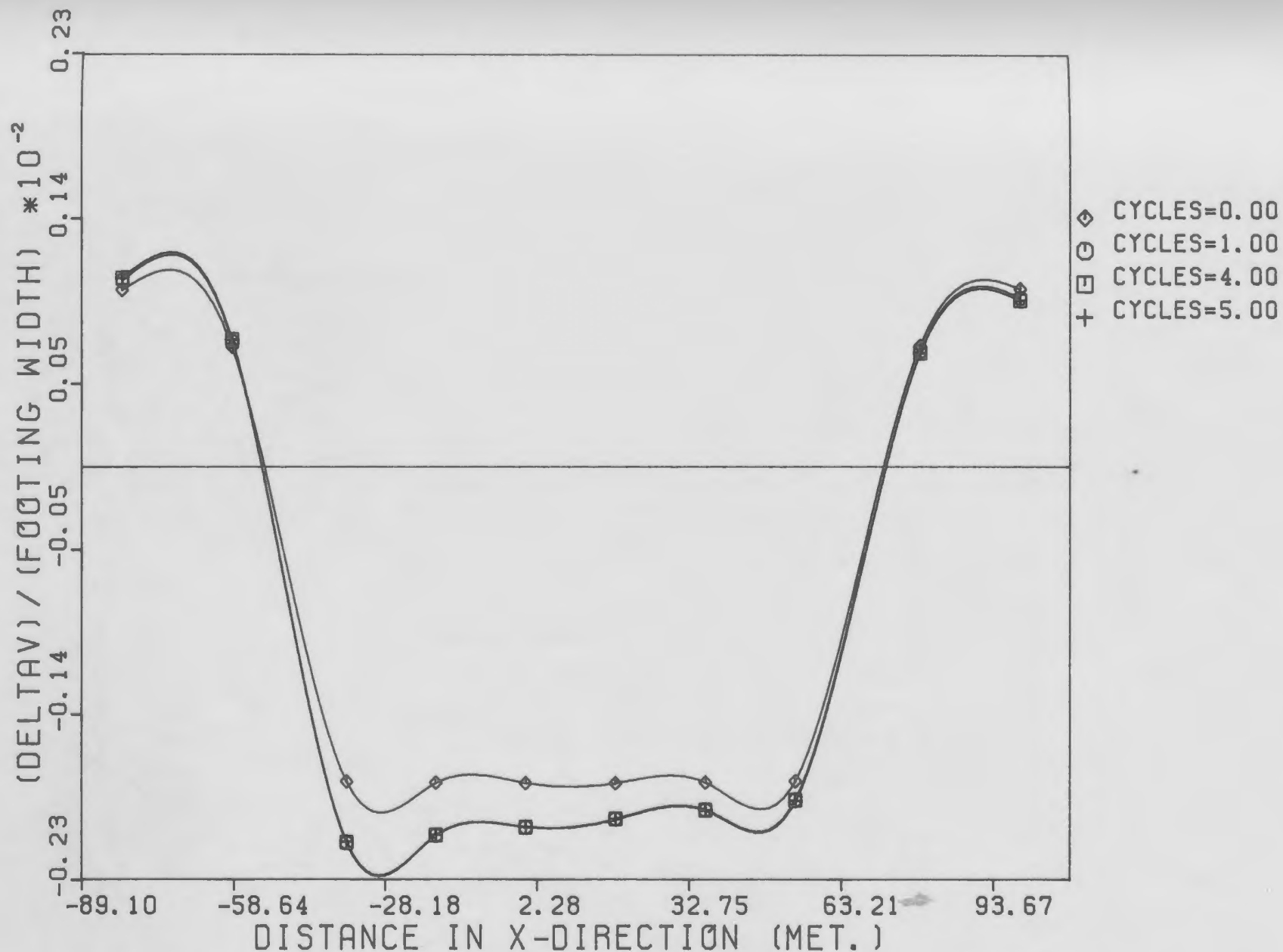


FIG. 6.21 (b) DISTRIBUTION OF VER. DISPLACEMENT
(WITH CAISSON; DYNAMIC ELASTIC ANALYSIS)
(NODES 16, 21, 27, 33, 39, 45, 51, 57, 62, 67 ETC)

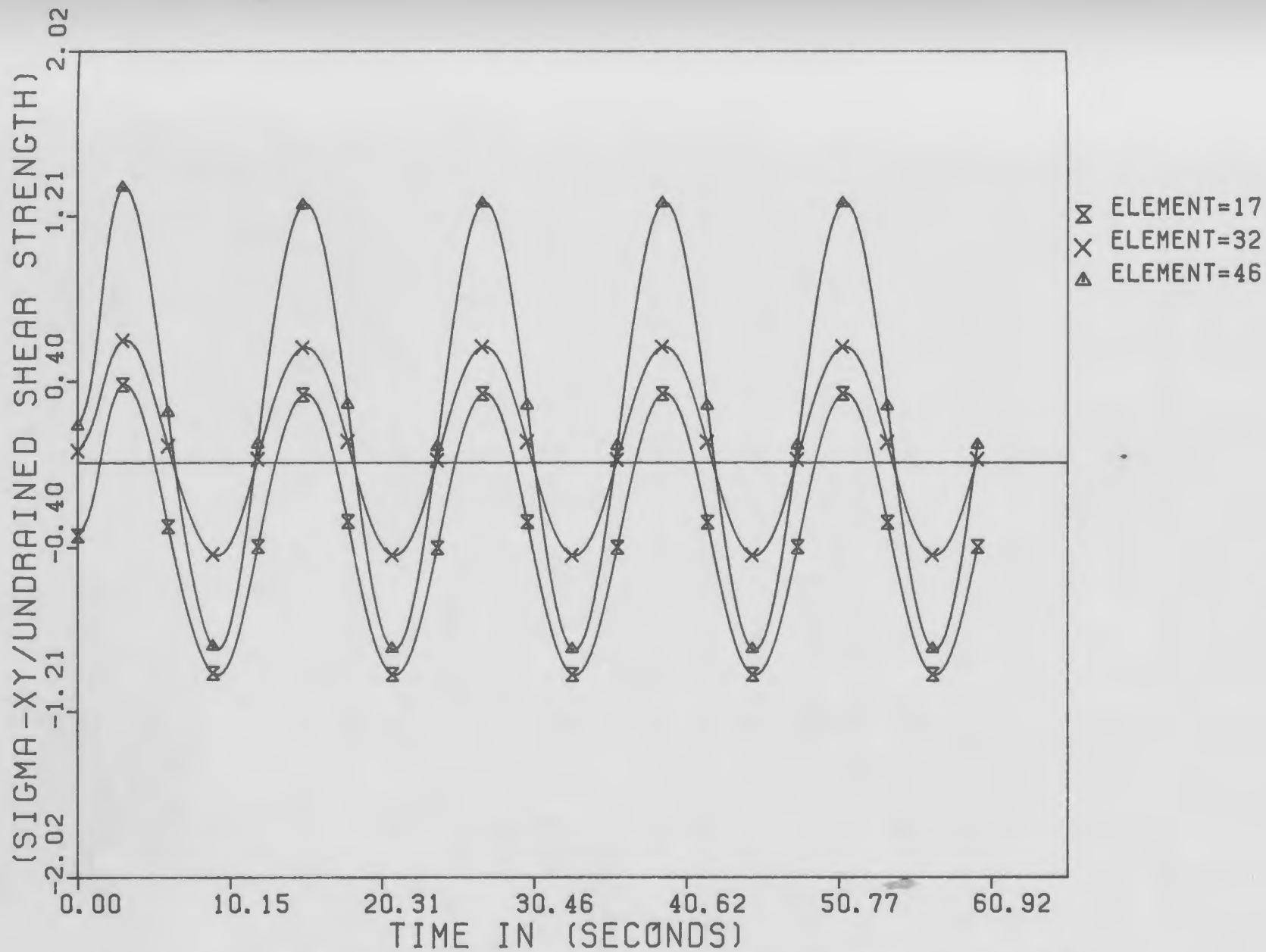


FIG. 6. 22 TIME HISTORY PLOT OF SHEAR STRESS
 (WITH CAISSON; DYNAMIC ELASTIC ANALYSIS)

1.15, the dynamic shakedown loading domain is obtained as $\frac{F_V}{As_u} = 2.33$ and $\pm \frac{F_H}{As_u} = 0.58$.

Figs. 6.20(a) and (b) depict the time history of elastic displacements in horizontal and vertical directions for selected nodal points. Because of the damping, steady state condition is reached within a very few cycles. The distributions of deformation δ_H and δ_V for various loading cycles are also shown in Figs. 6.21(a) and (b) respectively. Fig. 6.22 shows the shear stress time history for various elements beneath the caisson. The maximum shear stress is observed in element 46 because of the effect of the combined moment and shear loading acting at the right edge of the footing.

The horizontal and vertical loads are applied at top nodes of the caisson in a distributed manner and subsequently the dynamic response analyses are carried out with two types of nonlinear models: a) equivalent linear and b) elastic-plastic.

6.5.2 Equivalent Linear Analysis

The nonlinear effect of the soil due to the cyclic wave loading can be expressed in terms of the soil shear modulus, G , and the damping ratio, D , both

dependent on cyclic shear strains. In the equivalent linear analysis, the soil properties are adjusted by an iterative procedure so as to make them compatible with the computed soil strain level. The soil properties used in the analysis are the maximum shear modulus, G_{\max} , maximum damping ratio, D_{\max} , and undrained shear strength, s_u . For clays, the ratio of G_{\max}/s_u , which is constant (Fig. 5.3) is 2300 at low strain level ($10^{-4}\%$) and the maximum damping D_{\max} is 29% at 10% of strain (Fig. 5.4). For simplicity the secant shear modulus, G , versus the shear strain, and damping ratio are approximated by a series of straight lines on a semilogarithmic plot. These values are then stored in the programme and used to compute the strain compatible soil modulus and damping ratio during subsequent iterations.

Initial shear modulus is assigned to each element and the natural frequency of the structure-foundation system is first computed. The damping ratio, is selected corresponding to the first frequency of the system and the damping matrix of a typical soil element is formed from Eqn. (5.1). Having assembled the strain dependent damping matrix and the initial set of shear moduli for

each soil element, a step-by step direct integration is carried out following Eqn. (5.41) with a time step interval of 0.1429 second. The stress and strain histories obtained from the first analysis is examined for each element and the maximum shear strain is estimated. The new values of shear moduli and damping ratios corresponding to these shear strain amplitudes are estimated from Fig. 5.4 and Table 5.1.

The natural frequency of the soil-foundation system, ω_1 , is computed based on new shear moduli and the Rayleigh damping coefficients, B_1 and B_2 are then computed from Eqns. (5.2a) and (5.2b). The global damping matrix is formed again and the response analysis carried out for total number of 420 time steps in order to trace the five cycles of wave loading.

Iteration is continued until the difference between the actual modulus and damping used and the strain compatible modulus and damping values is within 10% or less, and the results obtained from the final iteration are treated as the nonlinear response. Table 6.8 and 6.9 show the converged values of the shear moduli and damping ratios for selected elements beneath the gravity foundation.

TABLE 6.8 - CONVERGENCE OF SOIL SHEAR
MODULUS, G, (kPa)

Element Numbers	Iteration Numbers			
	1	2	3	4
2	14651	17680	19425	20293
15	11900	12720	13073	13279
17	7791	7380	7430	7468
27	10800	10730	10712	10702
33	13563	14611	15320	15542
46	7673	7233	7213	7311

TABLE 6.9 - CONVERGENCE OF SOIL
DAMPING RATIO'S, D.

Element Numbers	Iteration Numbers			
	1	2	3	4
2	0.1337	0.1235	0.1175	0.1142
15	0.1515	.1441	.1410	0.1392
17	0.1886	0.1965	.1971	0.1915
27	0.1615	0.1621	0.1624	0.1623
33	0.1375	0.1339	0.1319	0.1307
46	0.1896	0.1936	0.1938	0.1930

Figs. 6.23(a) and (b) show the time histories of horizontal and vertical deformations for selected nodes beneath the caisson. Figs. 6.24(a) and (b) depict the distribution of δ_H and δ_V along horizontal directions for different loading cycles. Time history of shear stress is shown in Fig. 6.25 for selected elements beneath the caisson.

6.5.3 Dynamic Elastic-Plastic Analysis

In the nonlinear dynamic analysis, the incremental approach is used and the equilibrium iteration carried out within each time step of increment as described in Sections 5.2.3 and 5.3.2.

Figs. 6.26(a) and (b) show the time histories of nodal displacements in horizontal and vertical directions for selected nodal points beneath the caisson. Figs. 6.27(a) and 6.27(b) depict the distribution of δ_H and δ_V for different load cycles in horizontal directions. It is seen from these figures that the shakedown has taken place beneath the caisson after second loading cycle i.e. no further increment of displacement with respect to cyclic loading. Fig. 6.28 depicts the shear stress time history along the horizontal direction for the elements beneath the caisson structure.

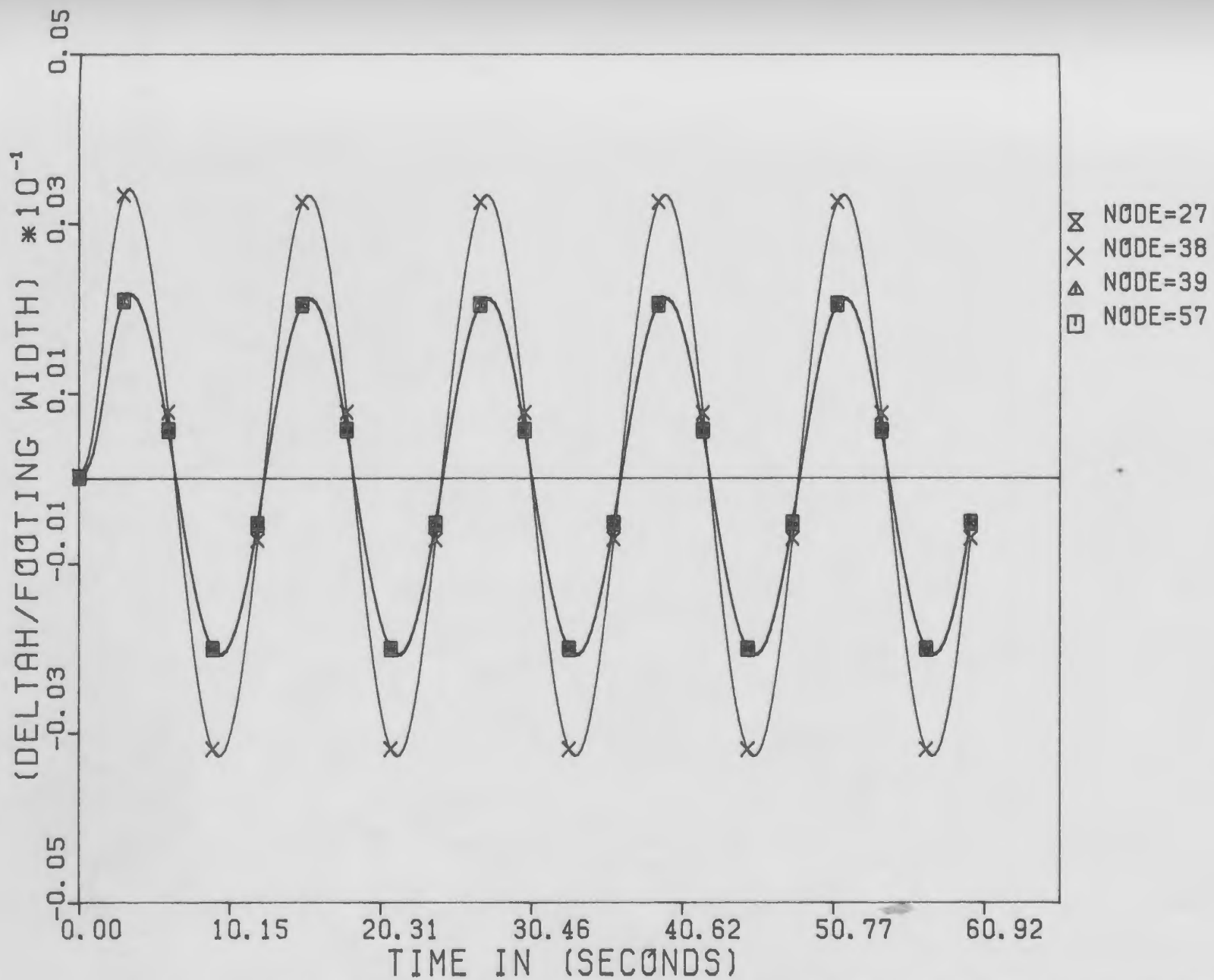


FIG. 6.23 (a) TIME HISTORY PLOT OF HOR. DISPLACEMENT (WITH CRISSON; EQUIVALENT LINEAR ANALYSIS)

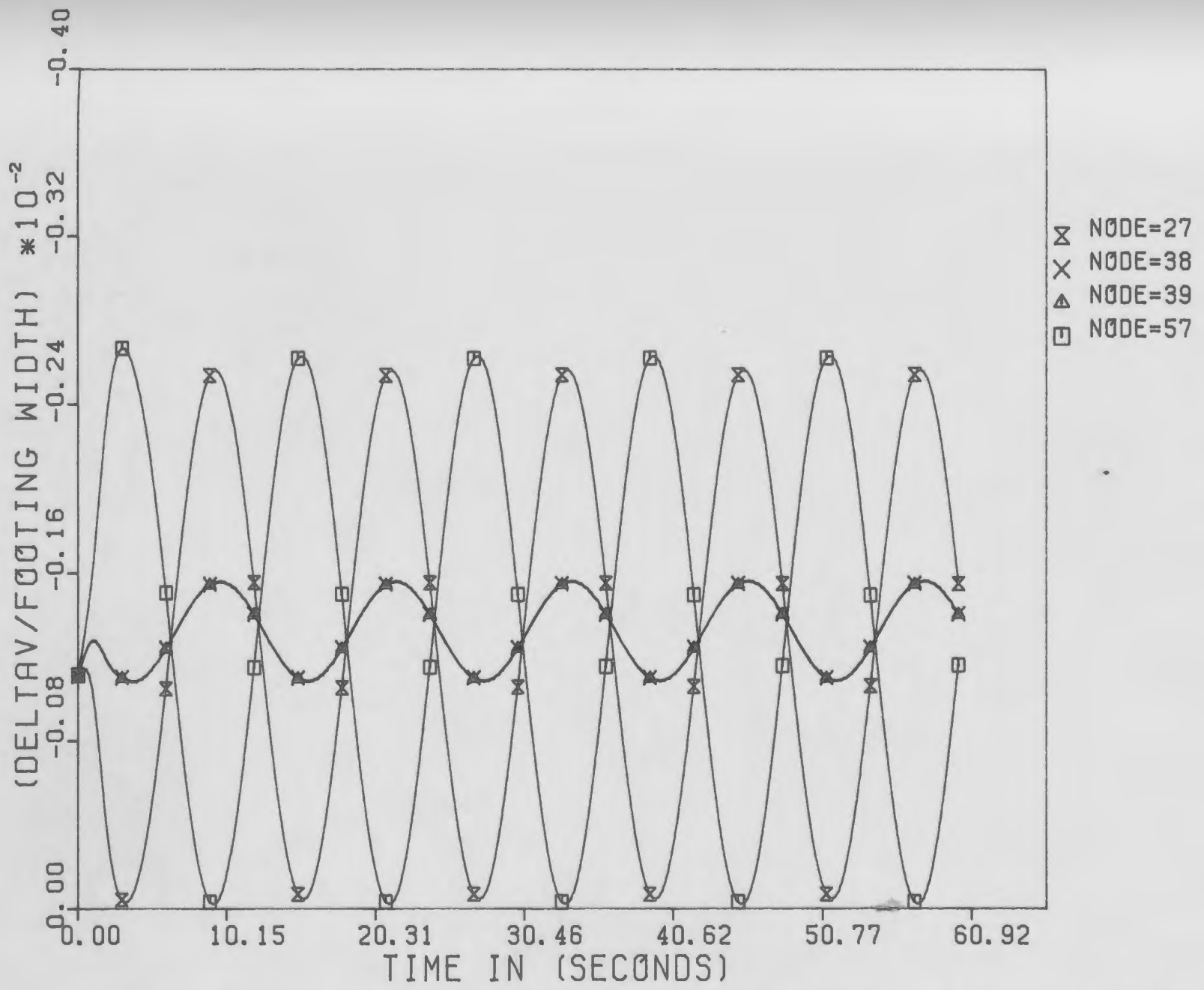


FIG. 6.23 (b) TIME HISTORY PLOT OF VER. DISPLACEMENT
(WITH CRISSON; EQUIVALENT LINEAR ANALYSIS)

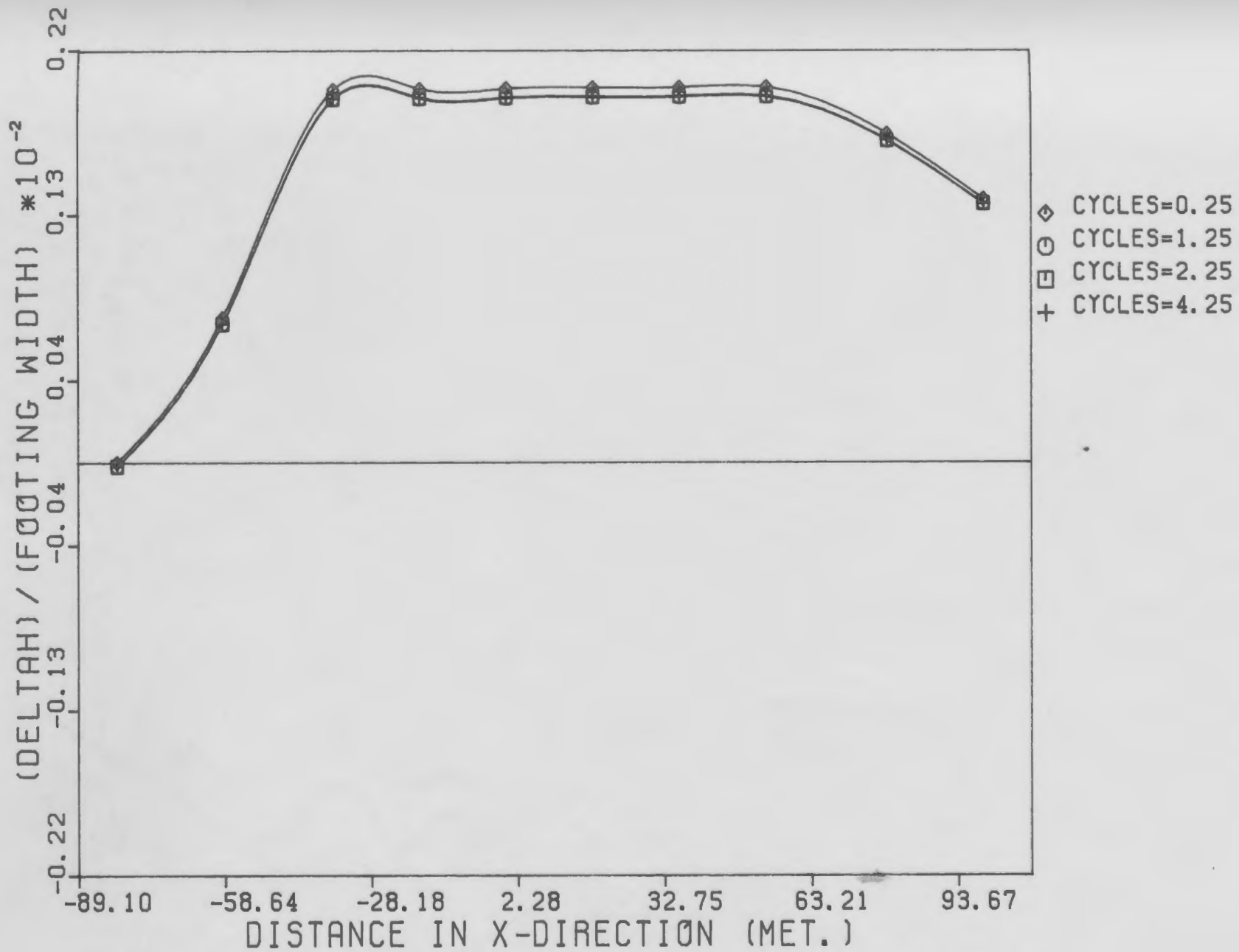


FIG. 6.24 (a) DISTRIBUTION OF HOR. DISPLACEMENT
(WITH CAISSON; EQUIVALENT LINEAR ANALYSIS)
(NODES 16, 21, 27, 33, 39, 45, 51, 57, 62, 67 ETC;)

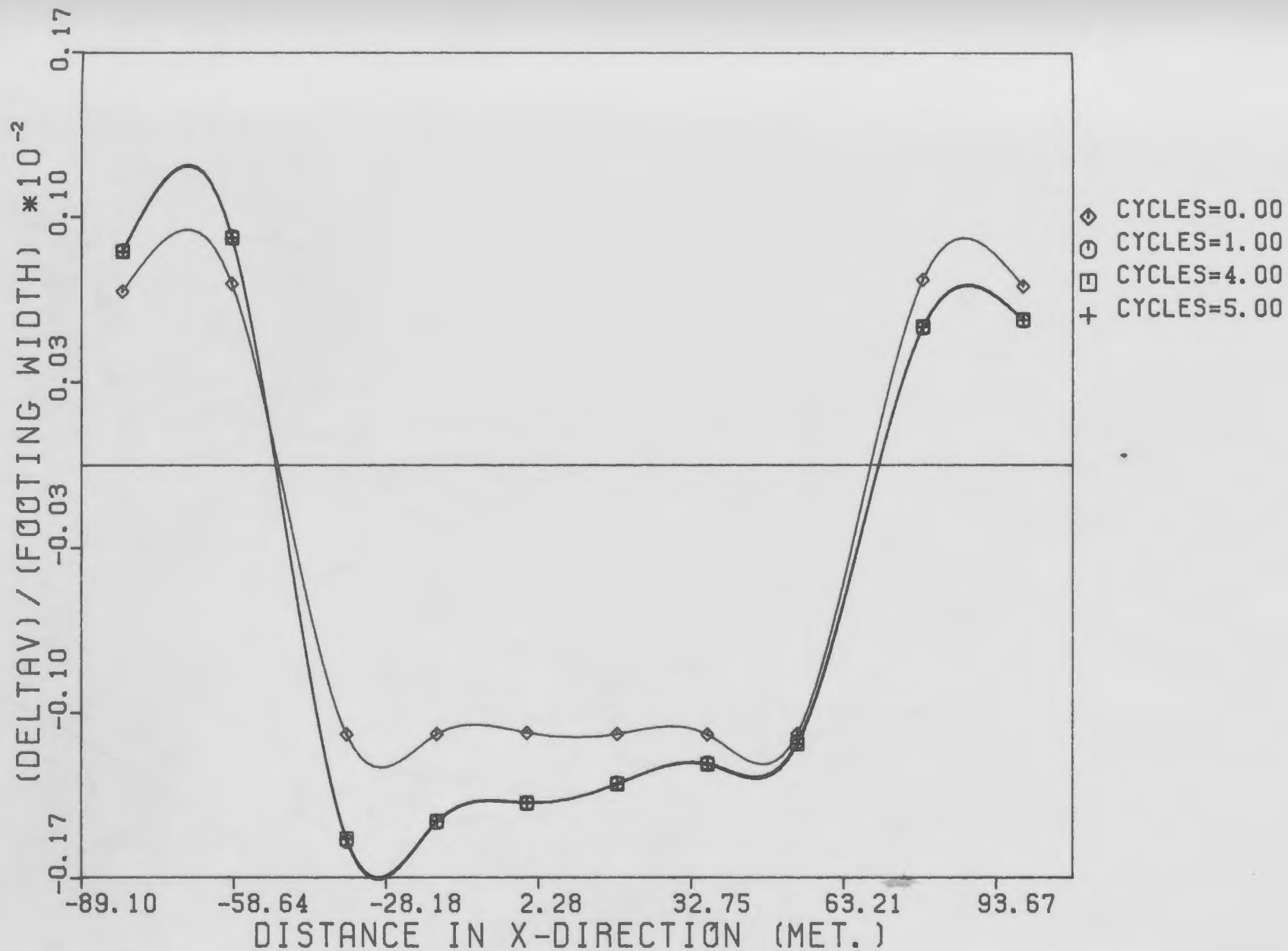


FIG. 6.24 (b) DISTRIBUTION OF VER. DISPLACEMENT
 (WITH CRISSON; EQUIVALENT LINEAR ANALYSIS)
 (NODES 16, 21, 27, 33, 39, 45, 51, 57, 62, 67 ETC;)

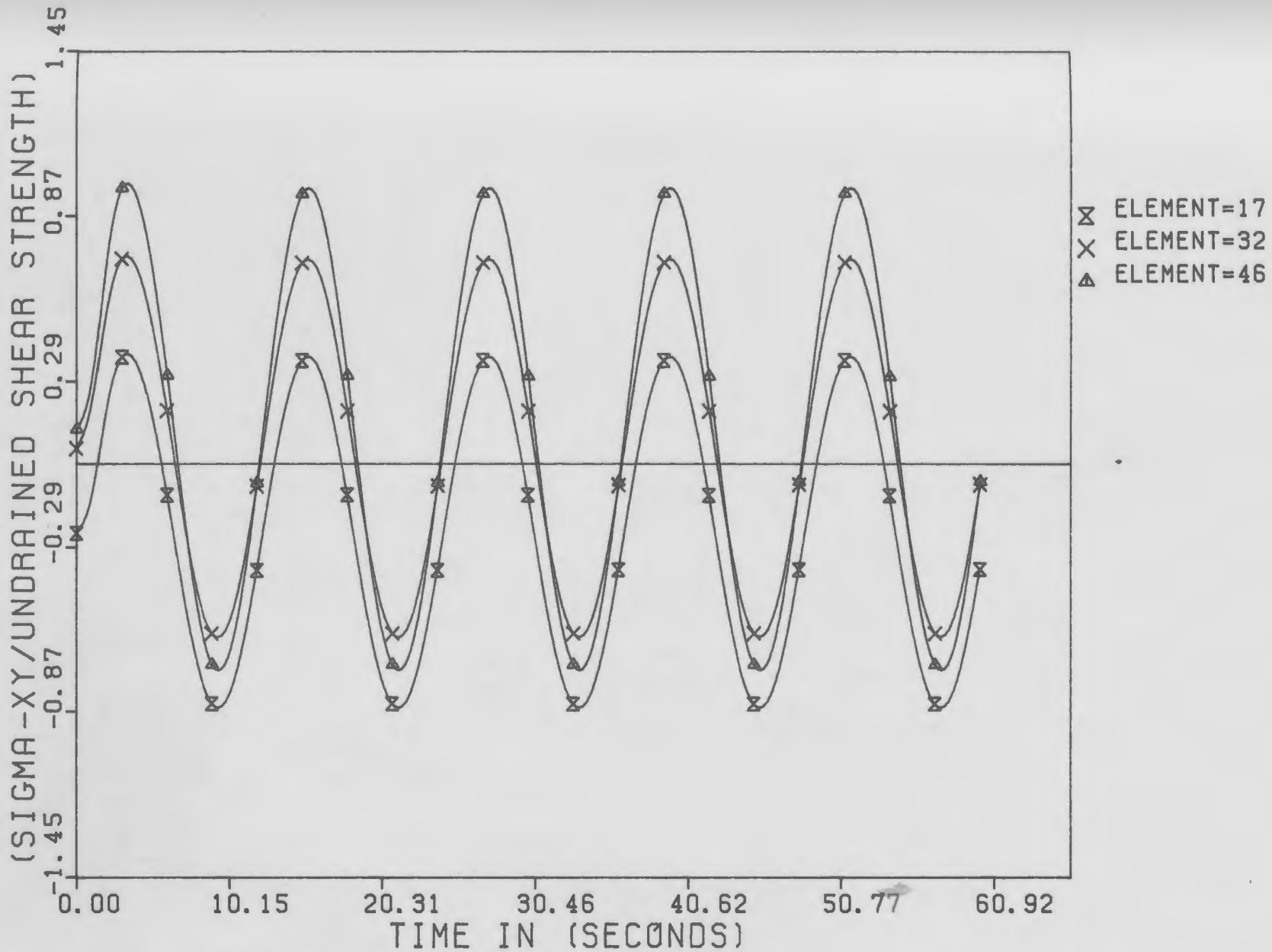


FIG. 6.25 TIME HISTORY PLOT OF SHEAR STRESS
(WITH CAISSON; EQUIVALENT LINEAR ANALYSIS)

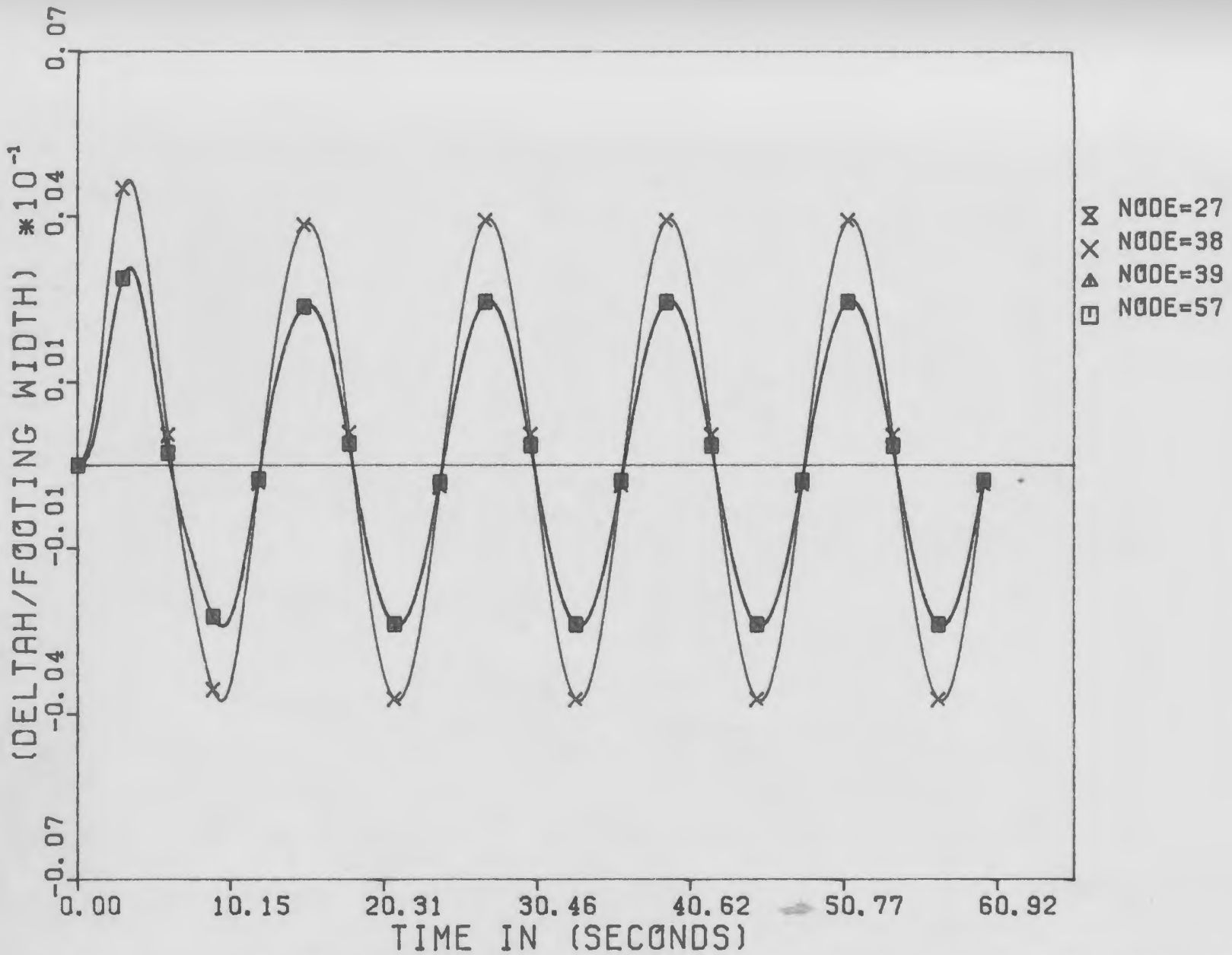


FIG. 6.26 (a) TIME HISTORY PLOT OF HOR. DISPLACEMENT
 (WITH CAISSON; DYNAMIC ELASTIC-
 PLASTIC ANALYSIS)

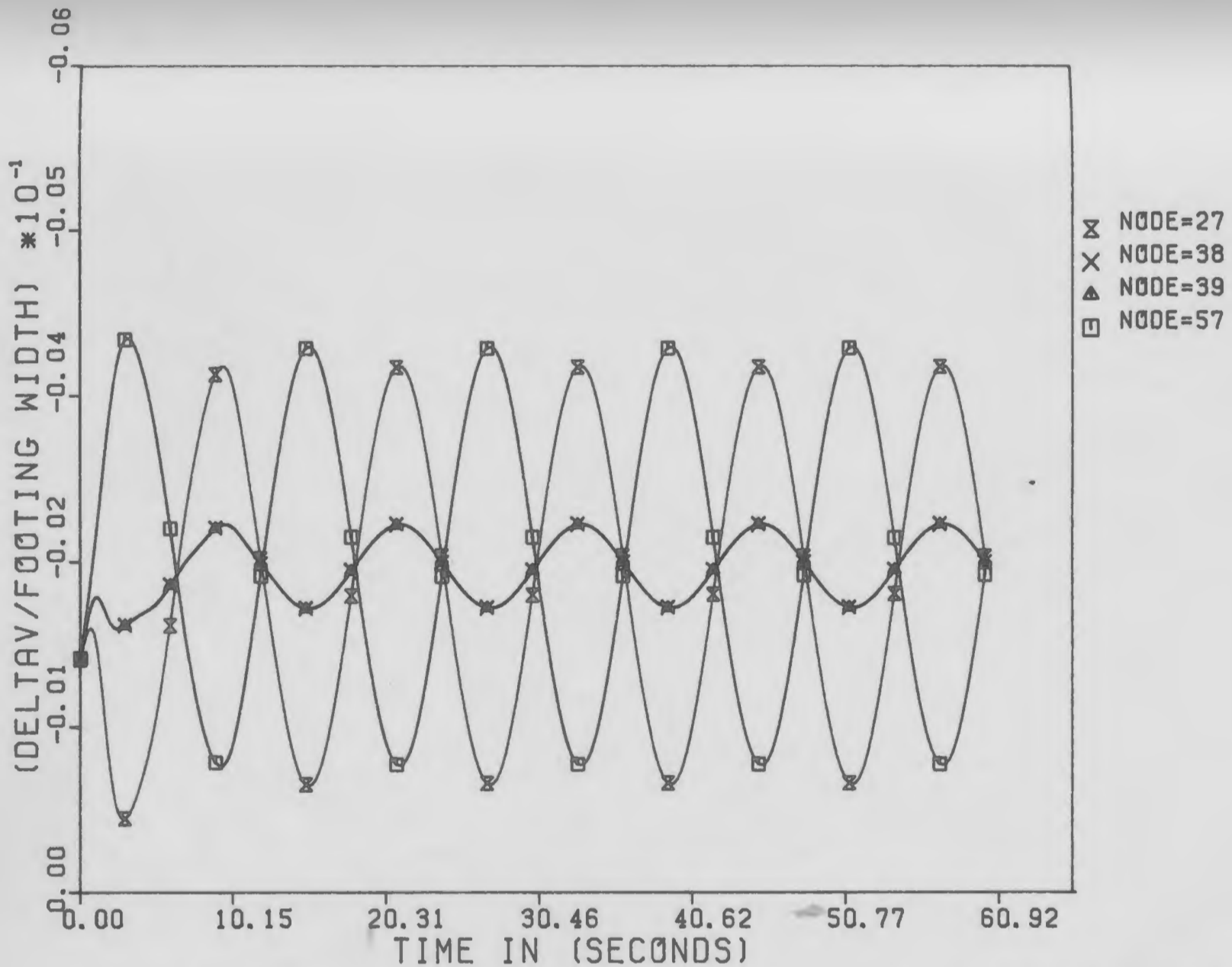


FIG. 6.26 (b) TIME HISTORY PLOT OF VER. DISPLACEMENT
 (WITH CRISSON; DYNAMIC ELASTIC PLASTIC ANALYSIS)

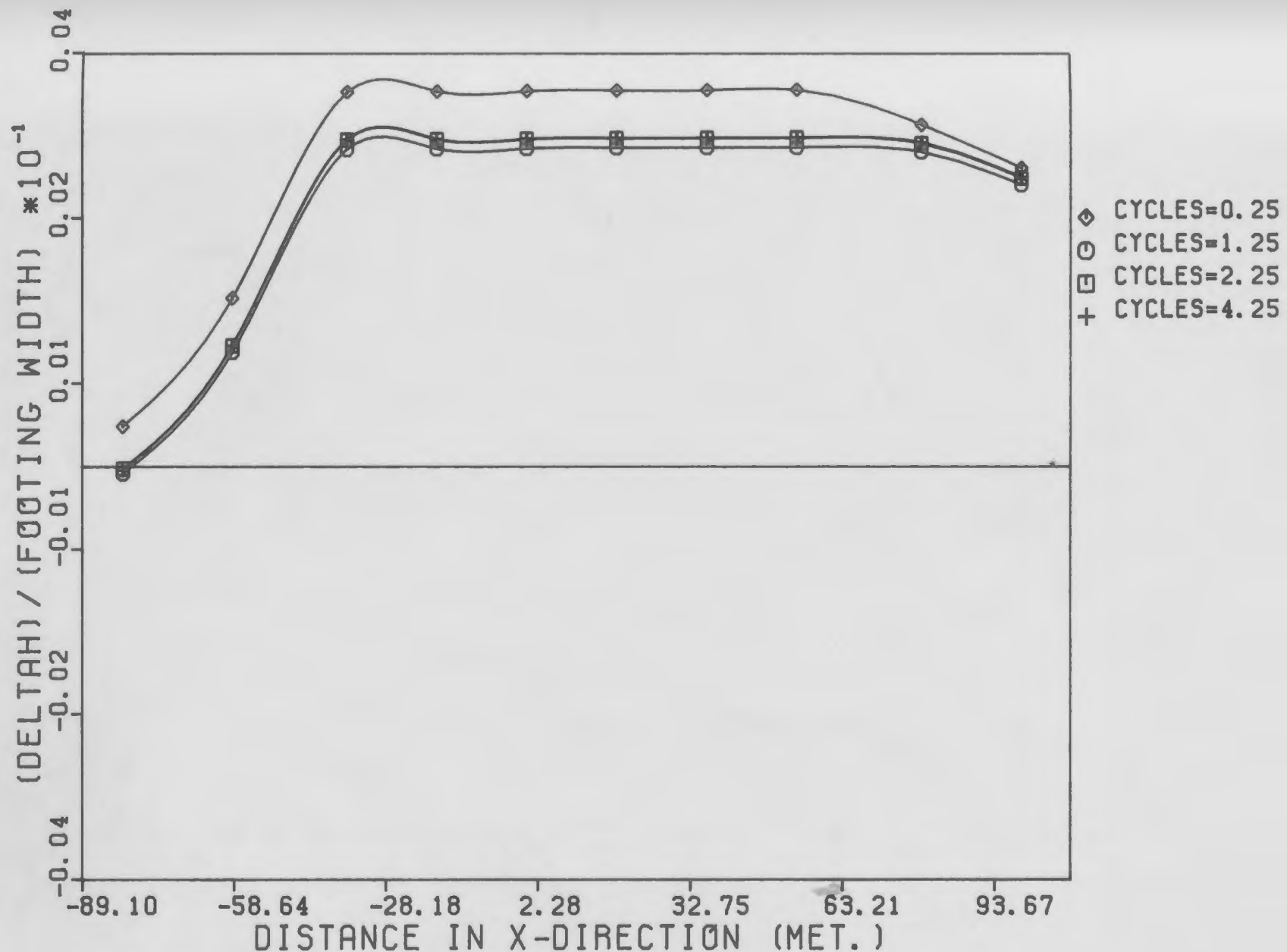


FIG. 6.27(a) DISTRIBUTION OF HOR. DISPLACEMENT
(WITH CAISSON; DYNAMIC PLASTIC ANALYSIS)
(NODES 16, 21, 27, 33, 39, 45, 51, 57, 62, 67 ETC;)

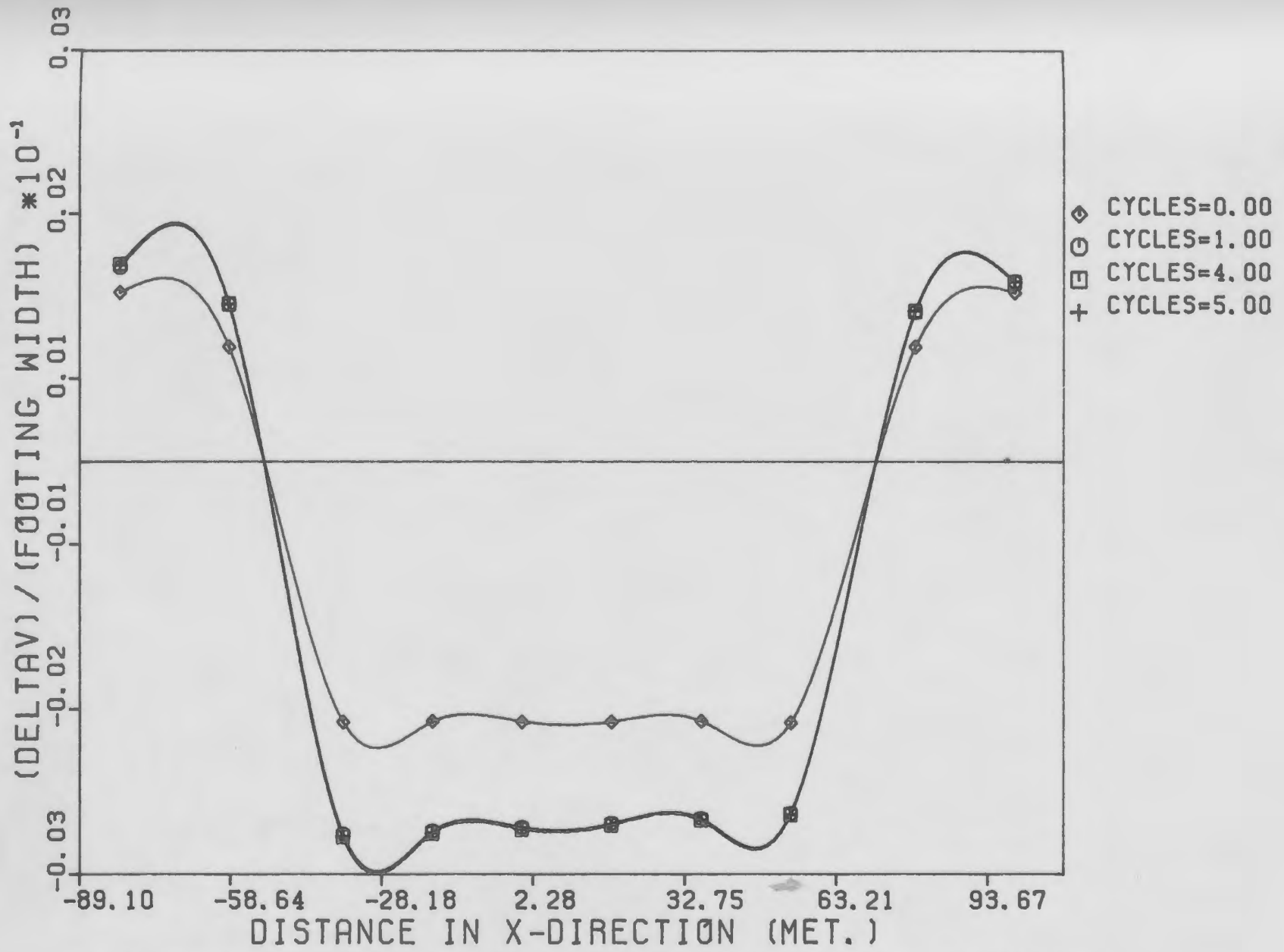


FIG. 6. 27 (b) DISTRIBUTION OF VER. DISPLACEMENT
(WITH CRISSON; DYNAMIC PLASTIC ANALYSIS)
(NODES 16, 21, 27, 33, 39, 45, 51, 57, 62, 67 ETC;)

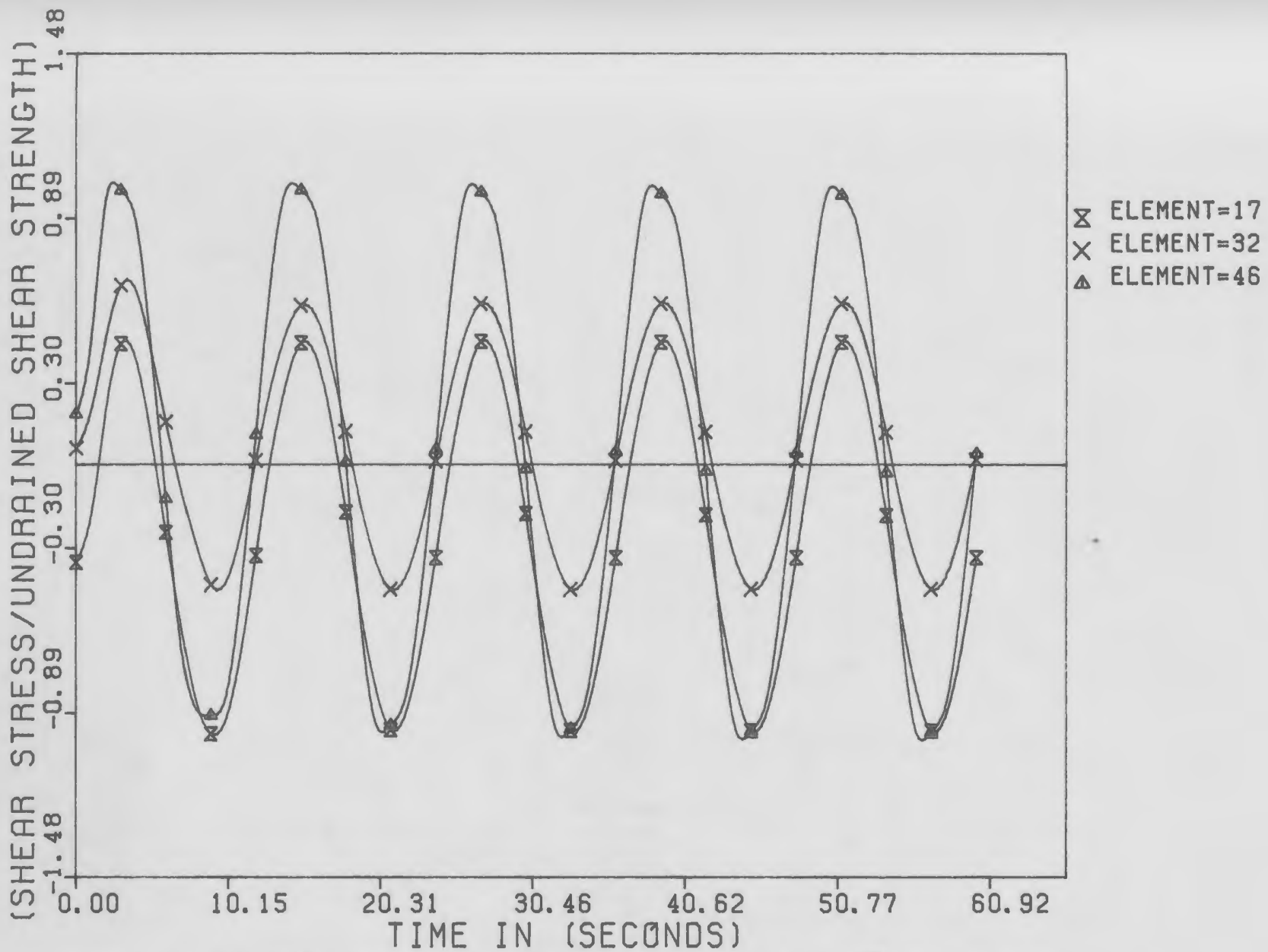


FIG. 6.28 TIME HISTORY PLOT OF SHEAR STRESS
(WITH CAISSON; DYNAMIC PLASTIC ANALYSIS)

6.5.4 Comparison Of Various Analyses

Four types of analysis have been carried out in previous sections. These analyses are: a) static non-linear b) dynamic elastic c) equivalent linear and d) dynamic elastic-plastic. Fig. 6.29 depicts the comparison of time histories of vertical displacement for node 38 for various analyses.

The dynamic elastic-plastic response is almost 8% lower than compared to its static nonlinear counterpart. However, the equivalent linear analysis suppresses the response as much as 40% compared to dynamic elastic-plastic analysis. The difference in the result can be attributed to the fact that in equivalent linear analysis the response is computed with the constant set of strain compatible soil properties for the entire loading history whereas in the true non-linear model, stiffness is updated in each time step based on the revised stress-state in each element. The limitation of the equivalent linear analysis is that it does not allow the computation of the permanent deformation.

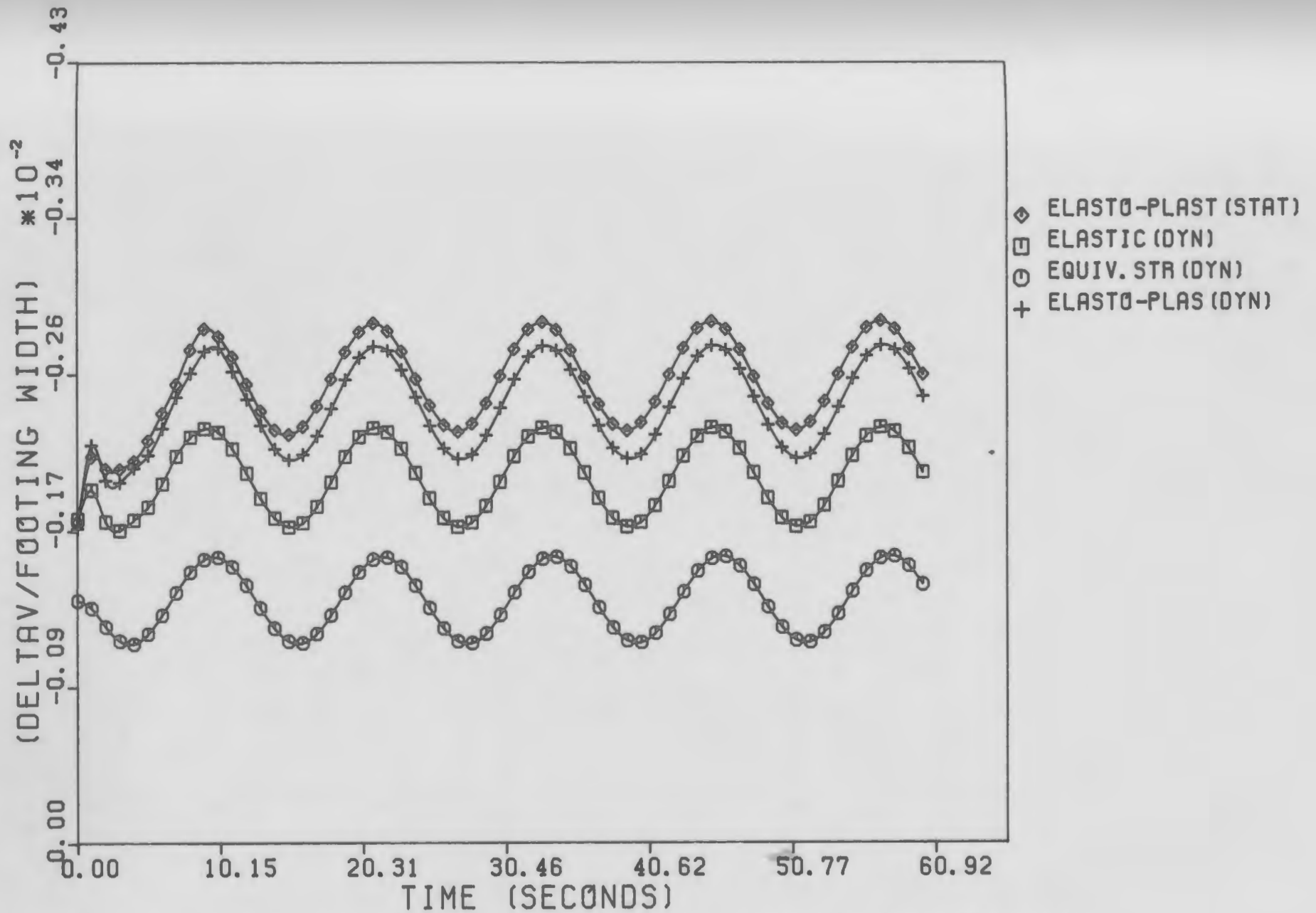


FIG. 6. 29 TIME HISRTORY PLOT OF VER. DISPLACEMENT
 (COMPARISON OF DIFFERENT SOIL MODELS)
 (WITH CAISSON; NODE 38)

CHAPTER VII

CONCLUSIONS7.1 Introduction

An analytical method is developed for the prediction of the permanent displacement of the foundation of a gravity type offshore platform resting on saturated soils. The wave loading is considered as dynamic and cyclic in nature and the soil is assumed to behave in an undrained manner.

The method is based on the finite element dynamic shakedown formulation of the saturated porous medium using a piecewise linearized convex yield surface. The dynamic shakedown load factor is obtained by first computing the elastodynamic response of the caisson-foundation (saturated) medium for some adjusted initial conditions (e.g. displacement and velocity) and then solving the problem as a linear programming one.

Conclusions

The displacement based finite element formulation of the saturated porous medium is used to analyze a strip footing problem for drained and undrained conditions. Analyses indicate that the displacement based finite element formulation can quite adequately predict the 'nearly incompressible' undrained deformation as well as the fully drained condition. Results show that the drained deformation is higher compared to its undrained counterpart. This is due to the fact that in undrained deformation volumetric deformation is almost zero and therefore additional restraint is imposed.

The dual form of the kinematic shakedown problem is used to compute the shakedown load factors for two types of problems. These are a) a thick cylinder under variable internal pressure and b) a strip footing under repeated vertical loading. In the cylinder problem three types of meshes are used and the analysis indicates that for a cylinder with 45 elements, elastic and shakedown limit pressures are in excellent agreement with the theoretical values. For the strip footing problem, vertical pressure at shakedown agrees well with the ultimate bearing capacity of the foundation.

Having computed the shakedown limit load for the strip footing, an incremental analysis is carried out for five cycles of loading. The vertical pressure is assumed to be triangular in shape and varies between zero and pressure at shakedown. Shakedown is observed within 3 cycles of the loading and the permanent displacement is computed as 0.82% of the footing width. The total displacement at shakedown load is 1.58% of the footing width.

The computations of elastic and shakedown limit pressures for a flexible foundation indicate that elastic and shakedown limit pressures depend on the inclination angle and eccentricity of the load. The elastic and shakedown limit pressures decrease as the eccentricity and the inclination angle increase. For example, the shakedown limit pressure can decrease as much as 54% (for an eccentricity of $e/B = 0.16$ and $\alpha_A = 10^\circ$) with respect to the case where the loading is purely vertical (i.e. $e/B = 0$ and $\alpha_A = 0.0$). Also, shakedown analyses indicate that vertical pressures under various eccentric loadings are below those predicted by the semi-empirical bearing capacity formula. Therefore, the use of the empirical formula will not produce a conservative estimate of the

bearing capacity of the foundation when the loading is cyclic in nature.

The frequencies for the undrained condition are higher than those obtained for drained situation. This is because of the high stiffness associated with the undrained condition.

Dynamic shakedown load factor for the stiff footing problem is only 10% lower than its static counterpart. This is because the amplification of stresses has occurred at frequency ratio of $\omega^* = 0.34$ and therefore the shakedown load factor has further decreased.

In computing the response analyses for three different soil models, it is observed that the equivalent linear analysis predicts lower deformation as opposed to the other two types of analyses. However, for the quasistatic analysis of the stiff footing problem, the permanent vertical deformation is within 0.085 percent of the footing width. In all cases, it is seen that shakedown has taken place with respect to the deformations of the foundation.

These findings of this present research lead to the following conclusions.

1. The vertical undrained deformation of a strip footing is almost 44% of its drained counterpart.
2. Shakedown limit pressures for a thick cylinder and a strip footing agree quite well with the theoretical values.
3. Shakedown limit pressures for a flexible footing decrease as the inclination angle and the eccentricity of the load increase.
4. The undrained natural frequency of the soil-foundation is 12% higher than its drained counterpart.
5. Dynamic shakedown limit load for the stiff footing problem is only 10% lower than its quasistatic counterpart.
6. Permanent vertical displacement of the foundation is within 0.08% of the footing width.

7. Equivalent linear model suppresses the foundation displacement as much as 40% than those compared with dynamic elastic-plastic analysis.

7.3

Contributions

1. Development of an analytical method in determining the permanent deformation of the foundation of a gravity base offshore foundation considering soil shakedown.
2. Computations of static and dynamic shakedown load factors for various types of problems e.g. gravity foundation under cyclic inclined eccentric loading conditions.
3. Computations of the response quantities for various types of soil models e.g. equivalent linear and elastic-perfectly plastic.
4. Development of a computer code OPFA (Offshore Platform Foundation Analysis) for carrying out the above computations.

7.4

Recommendations for Future Research

The following recommendations are made to extend the present work.

- Use of nonassociated flow rule so that the work softening criteria can be incorporated in the soil constitutive relationship.
- Geometric nonlinearity to be included in the shakedown analysis.
- Use of workhardening type soil model such as proposed by Prevost (1977) and Mroz et al (1978).
- Efficient algorithm for the solution of linear programming problem related to shakedown analysis.

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APPENDIX - A
PROOFS OF SHAKEDOWN THEOREMS

A1. PROOF OF MELAN'S THEOREM

For an elastic-perfectly plastic body experiencing simultaneously the action of a system of loads varying slowly in time within prescribed limits, $\sigma_{ij}^E(\bar{x}, t)$ and $\epsilon_{ij}^E(\bar{x}, t)$ are assumed as the instantaneous values of soil stresses and strains respectively in the corresponding elastic medium. $\sigma_{ij}(\bar{x}, t)$ and $\epsilon_{ij}(\bar{x}, t)$ denote the instantaneous values of the soil stresses and strains in the actual elastic-plastic state.

Therefore, $\sigma_{ij}^R(\bar{x}, t)$ and $\epsilon_{ij}^R(\bar{x}, t)$ can be expressed as

$$\sigma_{ij}^R(\bar{x}, t) = \sigma_{ij}(\bar{x}, t) - \sigma_{ij}^E(\bar{x}, t) \quad \dots (1)$$

and

$$\epsilon_{ij}^R(\bar{x}, t) = \epsilon_{ij}(\bar{x}, t) - \epsilon_{ij}^E(\bar{x}, t) \quad \dots (2)$$

where $\sigma_{ij}^R(\bar{x}, t)$ and $\epsilon_{ij}^R(\bar{x}, t)$ are instantaneous residual soil stresses and strains respectively in the body. The elastic strains which are caused by the residual soil stresses are denoted by ϵ_{ij}^{ER} . Strains, ϵ_{ij}^E , ϵ_{ij} are kinematically admissible, which means they satisfy the compatibility condition

and thus the corresponding displacements satisfy prescribed kinematic boundary conditions shown in Fig. 3.1.

The total soil strains, ϵ_{ij} , are composed of both elastic, ϵ_{ij}^E , and plastic, ϵ_{ij}^P , parts as:

$$\epsilon_{ij}(\bar{x}, t) = \epsilon_{ij}^E(\bar{x}, t) + \epsilon_{ij}^P(\bar{x}, t) \quad \dots (3)$$

Dividing Eqn. (1) by the appropriate elasticity constants we obtain

$$\epsilon_{ij}^{ER}(\bar{x}, t) = C_{ijkl}^{-1} \sigma_{ij}(\bar{x}, t) - \epsilon_{ij}^E(\bar{x}, t) \quad \dots (4)$$

where C_{ijkl}^{-1} = coefficients of elasticity matrix

Adding Eqn. (2) and Eqn. (4) and using Eqn. (3)

$$\epsilon_{ij}^R(\bar{x}, t) + C_{ijkl}^{-1} \sigma_{ij}(\bar{x}, t) = \epsilon_{ij}^{ER}(\bar{x}, t) + \epsilon_{ij}^P(\bar{x}, t) + \epsilon_{ij}^E(\bar{x}, t) \quad \dots (5)$$

Therefore

$$\epsilon_{ij}^R(\bar{x}, t) = \epsilon_{ij}^{ER}(\bar{x}, t) + \epsilon_{ij}^P(\bar{x}, t) \quad \dots (6)$$

and

and

$$\begin{aligned}\varepsilon_{ij}(\bar{x}, t) &= \varepsilon_{ij}^R(\bar{x}, t) + \varepsilon_{ij}^E(\bar{x}, t) \\ &= \varepsilon_{ij}^{ER}(\bar{x}, t) + \varepsilon_{ij}^P(\bar{x}, t) + \varepsilon_{ij}^E(\bar{x}, t) \dots (7)\end{aligned}$$

The admissible soil plastic strain rate cycle, $\dot{\varepsilon}_{ij}^P$, is defined by its property that the increments of the plastic strains, $\Delta\varepsilon_{ij}^R$, in such a cycle for some time interval, τ , given by

$$\Delta\varepsilon_{ij}^R = \int_0^\tau \dot{\varepsilon}_{ij}^P(t) dt \dots (8)$$

form a kinematically admissible strain distribution.

The strains given by Eqn. (8) may be obtained from Eqn. (3.7a) by prescribing a displacement field, Δu_i , which vanishes on S_1 , (Fig. 3.1). The soil plastic strain rate, $\dot{\varepsilon}_{ij}^P$, is related to residual stress rate distribution, $\dot{\sigma}_{ij}^R(\bar{x}, t)$. By definition, $\varepsilon_{ij}^{ER}(\bar{x}, t)$, corresponds to instantaneous elastic strain distribution caused by the soil residual stress state, $\sigma_{ij}^R(\bar{x}, t)$, and defining the, $\dot{u}_i(t)$, as the velocity field the kinematically admissible strain field is obtained from Eqn. (3.5) as

$$\dot{\varepsilon}_{ij} = \dot{\varepsilon}_{ij}^P + \dot{\varepsilon}_{ij}^{ER} \dots (9)$$

The displacement increment corresponding to the

admissible plastic strain rate cycle is

$$\Delta u_i = \int_0^{\tau} \dot{u}_i dt \quad \dots (10)$$

The residual soil stress at time $t = \tau$ will be equal to the value of the stress at time $t = 0$ as the plastic strain is kinematically admissible and hence

$$\int_0^{\tau} \dot{\epsilon}_{ij}^{ER} dt = 0 \quad \dots (11)$$

According to Melan's theorem if shakedown takes place, a time-independent residual stress field, $\bar{\sigma}_{ij}^R(\bar{x})$, will exist. This residual stress field, $\bar{\sigma}_{ij}^R(\bar{x})$, can be obtained from the solution of the equilibrium Eqn. (3.1) by satisfying the zero boundary conditions on S_2 , (Fig. 3.1) where the traction is prescribed. Defining, $\bar{\epsilon}_{ij}^{ER}$, as the elastic strain component corresponding to the fictitious residual stresses, $\bar{\sigma}_{ij}^{ER}(\bar{x})$, and admitting that, $\bar{\epsilon}_{ij}^{ER}$, are not kinematically possible strains, the following two stress fields are defined:

$$(1) \quad \sigma_{ij}^S = \sigma_{ij}^E + \bar{\sigma}_{ij}^R \quad \dots (12)$$

where σ_{ij}^S = safe soil stress state

and therefore $f_y(\sigma_{ij}^S) < 0$ from Eqn. (4.1a)

and

$$(2) \quad \sigma_{ij}^a = \sigma_{ij}^E + \sigma_{ij}^R \quad \dots (13)$$

where σ_{ij}^a = admissible soil stress which touches the yield surface and therefore satisfies the yield condition from Eqn. (4.1b) as

$$f_y(\sigma_{ij}^a) \leq 0$$

A fictitious positive elastic strain energy, S^e , is defined in terms of the stress difference, $(\sigma_{ij}^R - \bar{\sigma}_{ij}^R)$, where, $\bar{\sigma}_{ij}^R$, is a time independent residual stress field satisfying the equilibrium with zero external forces

Therefore

$$S^e = 1/2 \int_V (\sigma_{ij}^R - \bar{\sigma}_{ij}^R) (\epsilon_{ij}^{ER} - \bar{\epsilon}_{ij}^{ER}) dV \quad \dots (14)$$

The stress differences, $(\sigma_{ij}^R - \bar{\sigma}_{ij}^R)$, are related to strain differences, $(\epsilon_{ij}^{ER} - \bar{\epsilon}_{ij}^{ER})$, by Hooke's law.

Taking the derivative of the strain energy, S^e , with respect to time, we get

$$\dot{S}^e = \frac{d}{dt}(S^e) = \int_V (\sigma_{ij}^R - \bar{\sigma}_{ij}^R) \frac{d}{dt} (\epsilon_{ij}^{ER} - \bar{\epsilon}_{ij}^{ER}) dV \quad \dots (15)$$

As the stress, $\bar{\sigma}_{ij}^R$, and the strain, $\bar{\epsilon}_{ij}^{ER}$, are independent of time, by definition

$$\dot{S}^e = \int_V (\sigma_{ij}^R - \bar{\sigma}_{ij}^R) \dot{\epsilon}_{ij}^{ER} dV \quad \dots (16)$$

From Eqn. (7)

$$\dot{\epsilon}_{ij}^{ER} = \dot{\epsilon}_{ij} - \dot{\epsilon}_{ij}^E - \dot{\epsilon}_{ij}^p$$

Therefore

$$\dot{S}^e = \int_V (\sigma_{ij}^R - \bar{\sigma}_{ij}^R) (\dot{\epsilon}_{ij} - \dot{\epsilon}_{ij}^E - \dot{\epsilon}_{ij}^p) dV \quad \dots (17)$$

The distribution of the stress differences, $(\sigma_{ij}^R - \bar{\sigma}_{ij}^R)$, is self-equilibrating ie. it satisfies the equilibrium conditions with zero external force and the strain rate difference, $(\dot{\epsilon}_{ij} - \dot{\epsilon}_{ij}^E)$, is kinematically admissible. Therefore by principle of virtual work, it can be stated that the rate of work of the internal force(s) is equal to the rate of work of the external force(s). As the external forces are zero in this case, while, $u_i - u_i^E = 0$ on part of the surface, S_1 , (Fig. 3.1) where displacements are prescribed and using the following relationship

$$\int_V (\sigma_{ij}^R - \bar{\sigma}_{ij}^R) (\dot{\epsilon}_{ij} - \dot{\epsilon}_{ij}^E) dV = 0 \quad \dots (18)$$

Eqn. (17) is further reduced to

$$\dot{S}^e = \int_V (\sigma_{ij}^R - \bar{\sigma}_{ij}^R) \dot{\epsilon}_{ij}^p dV \quad \dots (19)$$

From Eqns. (12) and (13), Eqn. (19) can be written in the following form

$$\dot{S}^e = - \int_V (\sigma_{ij} - \sigma_{ij}^S) \dot{\epsilon}_{ij}^p dV \quad \dots (20)$$

The basic idea and proposition of Melan's theorem rest on the assumption that the material (in this case soil medium) obeys Drucker's postulate which is given by Eqn. (4.2a) as

$$(\sigma_{ij} - \sigma_{ij}^S) \dot{\epsilon}_{ij}^p > 0$$

Therefore, $\frac{d}{dt}(S^e) < 0$, as long as, $\dot{\epsilon}_{ij}^p = 0$; as the elastic strain energy is always positive and can never be negative, and therefore a time denoted by the passage of N cycles, will be reached when plastic flow will cease ie. $\dot{\epsilon}_{ij}^p = 0$ implying

$$\frac{d}{dt}(S^e) = 0 \quad \dots (21)$$

This means the residual stress field will no longer change with respect to time and the soil will subsequently experience only elastic deformation as the loads are varied cyclically.

A2. PROOF OF DYNAMIC SHAKEDOWN THEOREM

In order to prove the dynamic shakedown theorem as an extension of Melan's quasistatic shakedown theorem, an additional term is added to the Eqn. (14) which is referred to here as the kinetic energy of the system, K^e . The geometry change is assumed to be negligible.

$$\begin{aligned}
 TE &= \text{Total energy} \\
 &= S^e + K^e \\
 &= \int_V 1/2 (\sigma_{ij}^R - \bar{\sigma}_{ij}^R) (\epsilon_{ij}^{ER} - \bar{\epsilon}_{ij}^{ER}) dV \\
 &\quad + \int_V 1/2 m (\dot{u}_i - \dot{u}_i^E) (\dot{u}_i - \dot{u}_i^E) dV \\
 &\quad \dots (22)
 \end{aligned}$$

The virtual work equation for the dynamic loading situation can be written in the following rate form as

$$\int_S \hat{T}_i \dot{u}_i dS + \int_V B_i \dot{u}_i dV - \int_V m \ddot{u}_i \dot{u}_i dV = \int_V \sigma_{ij} \dot{\epsilon}_{ij} dV \dots (23)$$

By virtue of Eqn. (23), the derivative of Eqn. (22) with respect to time is expressed as

$$\begin{aligned}
\frac{d}{dt} (T^e) &= \int_V (\sigma^R - \bar{\sigma}^R) \frac{d}{dt} (\epsilon^{ER} - \bar{\epsilon}^{ER}) dV \\
&+ [\int_{S_2} (\hat{T}_i - \hat{T}_i^E) (\dot{u}_i - \dot{u}_i^E) dS + \int_V (B_i - B_i^E) \\
&(\dot{u}_i - \dot{u}_i^E) dV - \int_V (\sigma - \sigma^E) (\dot{\epsilon} - \dot{\epsilon}^E) dV] \\
&\dots (24)
\end{aligned}$$

As, $\bar{\epsilon}^{ER}$, is independent of time and the elastic solution, \dot{u}_i^E , must satisfy the same loading and constraints as given in Eqn. (3.7), both terms inside the bracket are zero. Using Eqn. (7),

$$\begin{aligned}
\dot{T}^e &= \int_V (\sigma^R - \bar{\sigma}^R) \dot{\epsilon}^{ER} dV - \int_V (\sigma - \sigma^E) (\dot{\epsilon} - \dot{\epsilon}^E) dV \\
&= \int_V (\sigma^R - \bar{\sigma}^R) (\dot{\epsilon} - \dot{\epsilon}^P - \dot{\epsilon}^E) dV \\
&- \int_V (\sigma - \sigma^E) (\dot{\epsilon} - \dot{\epsilon}^E) dV \dots (25)
\end{aligned}$$

Substituting $\bar{\sigma}^R = \sigma - \sigma^E$, in Eqn. (25) gives

$$\begin{aligned}
\dot{T}^e &= - \int_V (\sigma^R - \bar{\sigma}^R) \dot{\epsilon}^P dV - \int_V \bar{\sigma}^R (\dot{\epsilon} - \dot{\epsilon}^E) dV \\
&\dots (26)
\end{aligned}$$

The second term is again zero by the virtual work equation due to the fact that, $\bar{\sigma}^R$, is a time independent residual stress distribution and is always in equilibrium with zero external force.

Therefore

$$\dot{T}^e = - \int_V (\sigma^R - \bar{\sigma}^R) \dot{\epsilon}^P dV \quad \dots (27)$$

From Eqns. (12) and (13), Eqn. (27) reduces to

$$\dot{T}^e = - \int_V (\sigma - \sigma^S) \dot{\epsilon}^P dV \quad \dots (28)$$

$(\sigma - \sigma^S) \dot{\epsilon}^P > 0$, because of Drucker's stability postulate and therefore, $\dot{T}^e < 0$, as long as, $\dot{\epsilon}^P > 0$; the total energy is always positive and can never be negative, and therefore a time denoted by the passage of N cycles, will be reached when plastic flow will cease i.e. implying

$$\dot{T}^e = 0 \quad \dots (29)$$

This means the residual stress field will no longer change with respect to time under dynamic situation and the soil will subsequently experience only elastic deformation as the loads are varied cyclically.

APPENDIX - B
DERIVATION OF UNDRAINED
ELASTIC PLASTIC CONSTITUTIVE
MATRIX, EQN. (5.35)

B1. DERIVATION OF EQN. (5.35)

Using Eqn. (3.38h), Eqn. (5.26) and Eqn. (5.31), the term in the denominator in Eqn. (5.35) is expressed for elastic-perfectly plastic material ($H = 0$) as

$$\begin{aligned}
 [L] &= \left[a\delta_{ij} + \frac{S_{ij}}{2\sqrt{J_2}} \right]^T \begin{bmatrix} \lambda+2\mu & \lambda & \lambda & 0 \\ \lambda & \lambda+2\mu & \lambda & 0 \\ \lambda & \lambda & \lambda+2\mu & 0 \\ 0 & 0 & 0 & \mu \end{bmatrix} \left[a\delta_{ij} + \frac{S_{ij}}{2\sqrt{J_2}} \right] \dots (1) \\
 &= \left[a + \frac{S_x}{2\sqrt{J_2}} \quad a + \frac{S_y}{2\sqrt{J_2}} \quad a + \frac{S_z}{2\sqrt{J_2}} \quad \frac{\tau_{xy}}{\sqrt{J_2}} \right]^T \begin{bmatrix} \lambda+2\mu & \lambda & \lambda & 0 \\ \lambda & \lambda+2\mu & \lambda & 0 \\ \lambda & \lambda & \lambda+2\mu & 0 \\ 0 & 0 & 0 & \mu \end{bmatrix} \begin{bmatrix} a + \frac{S_x}{2\sqrt{J_2}} \\ a + \frac{S_y}{2\sqrt{J_2}} \\ a + \frac{S_z}{2\sqrt{J_2}} \\ \frac{\tau_{xy}}{\sqrt{J_2}} \end{bmatrix}
 \end{aligned}$$

$$\begin{aligned}
&= [(\lambda+2\mu) \left\{ \left(a + \frac{S_x}{2\sqrt{J_2}} \right)^2 + \left(a + \frac{S_y}{2\sqrt{J_2}} \right)^2 + \left(a + \frac{S_z}{2\sqrt{J_2}} \right)^2 \right\} \\
&\quad + 2\lambda \left\{ \left(a + \frac{S_x}{2\sqrt{J_2}} \right) \left(a + \frac{S_y}{2\sqrt{J_2}} \right) + \left(a + \frac{S_x}{2\sqrt{J_2}} \right) \left(a + \frac{S_z}{2\sqrt{J_2}} \right) \right. \\
&\quad \left. + \left(a + \frac{S_y}{2\sqrt{J_2}} \right) \left(a + \frac{S_z}{2\sqrt{J_2}} \right) \right\} + \mu \left(\frac{\tau_{xy}}{\sqrt{J_2}} \right)^2] \\
&= [(\lambda+2\mu) \left\{ 3a^2 + \frac{2a}{2\sqrt{J_2}} (S_x + S_y + S_z) + \frac{S_x^2 + S_y^2 + S_z^2}{4J_2} \right\} \\
&\quad + 2\lambda \left\{ 3a^2 + \frac{1}{4J_2} (S_x S_y + S_y S_z + S_z S_x) \right\} + \mu \frac{\tau_{xy}^2}{J_2}]
\end{aligned}$$

Since $S_x + S_y + S_z = 0$, one may write

$$\begin{aligned}
L &= [(\lambda+2\mu) (3a^2) + \frac{\lambda}{4J_2} \{ S_x^2 + S_y^2 + S_z^2 + 2(S_x S_y + S_y S_z + S_z S_x) \} \\
&\quad + 3a^2 (2\lambda) + \mu \frac{\tau_{xy}^2}{J_2} + \frac{\mu}{2J_2} (S_x^2 + S_y^2 + S_z^2)] \\
&= [3a^2 (3\lambda+2\mu) + \mu] \dots (2)
\end{aligned}$$

The each term of the numerator in Eqn. (5.35) is derived as follows

$$\begin{aligned}
\text{Assume } r_1 &= \frac{\partial f}{\partial \sigma_x} = a + \frac{S_x}{2\sqrt{J_2}} \\
r_2 &= \frac{\partial f}{\partial \sigma_y} = a + \frac{S_y}{2\sqrt{J_2}} \\
r_3 &= \frac{\partial f}{\partial \sigma_z} = a + \frac{S_z}{2\sqrt{J_2}}
\end{aligned} \dots (3)$$

$$\text{and } r_4 = \frac{\partial f}{\partial \tau_{xy}} = \frac{\tau_{xy}}{\sqrt{J_2}}$$

Using the above relationships, Eqn. (5.35) is expressed as

$$[D_{sd}^{Ep}] = [D_{sd}] - \frac{[D_{sd}] \begin{Bmatrix} \frac{\partial f_y}{\partial \sigma} \\ \frac{\partial f_y}{\partial \sigma} \end{Bmatrix} \begin{Bmatrix} \frac{\partial f_y}{\partial \sigma} \\ \frac{\partial f_y}{\partial \sigma} \end{Bmatrix}^T [D_{sd}]}{[L]} \dots (4)$$

$$= [D_{sd}] - [L]^{-1} \begin{bmatrix} \lambda+2\mu & \lambda & \lambda & 0 \\ \lambda & \lambda+2\mu & \lambda & 0 \\ \lambda & \lambda & \lambda+2\mu & 0 \\ 0 & 0 & 0 & \mu \end{bmatrix} \begin{bmatrix} r_1 \\ r_2 \\ r_3 \\ r_4 \end{bmatrix} \begin{bmatrix} r_1 & r_2 & r_3 & r_4 \end{bmatrix} \begin{bmatrix} \lambda+2\mu & \lambda & \lambda & 0 \\ \lambda & \lambda+2\mu & \lambda & 0 \\ \lambda & \lambda & \lambda+2\mu & 0 \\ 0 & 0 & 0 & \mu \end{bmatrix}$$

$$= [D_{sd}] - [L]^{-1} \begin{bmatrix} \lambda+2\mu & \lambda & \lambda & 0 \\ \lambda & \lambda+2\mu & \lambda & 0 \\ \lambda & \lambda & \lambda+2\mu & 0 \\ 0 & 0 & 0 & \mu \end{bmatrix} \begin{bmatrix} r_1^2 & r_1 r_2 & r_1 r_3 & r_1 r_4 \\ r_2 r_1 & r_2^2 & r_2 r_3 & r_2 r_4 \\ r_3 r_1 & r_3 r_2 & r_3^2 & r_3 r_4 \\ r_4 r_1 & r_4 r_2 & r_4 r_3 & r_4^2 \end{bmatrix} \begin{bmatrix} \lambda+2\mu & \lambda & \lambda & 0 \\ \lambda & \lambda+2\mu & \lambda & 0 \\ \lambda & \lambda & \lambda+2\mu & 0 \\ 0 & 0 & 0 & \mu \end{bmatrix} \dots (5)$$

Therefore,

$$\begin{aligned}
 [D_{sd}^{Ep}] &= [D_{sd}] - [L]^{-1} \begin{bmatrix} X_{11} & X_{12} & X_{13} & X_{14} \\ X_{21} & X_{22} & X_{23} & X_{24} \\ X_{31} & X_{32} & X_{33} & X_{34} \\ X_{41} & X_{42} & X_{43} & X_{44} \end{bmatrix} \begin{bmatrix} \lambda+2\mu & \lambda & \lambda & 0 \\ \lambda & \lambda+2\mu & \lambda & 0 \\ \lambda & \lambda & \lambda+2\mu & 0 \\ 0 & 0 & 0 & \mu \end{bmatrix} \\
 &= [D_{sd}] - [L]^{-1} \begin{bmatrix} Y_{11} & Y_{12} & Y_{13} & Y_{14} \\ Y_{21} & Y_{22} & Y_{23} & Y_{24} \\ Y_{31} & Y_{32} & Y_{33} & Y_{34} \\ Y_{41} & Y_{42} & Y_{42} & Y_{44} \end{bmatrix} \dots (6)
 \end{aligned}$$

$$\text{where } Y_{11} = (\lambda+2\mu)X_{11} + \lambda X_{12} + \lambda X_{13}$$

$$Y_{12} = \lambda X_{11} + (\lambda+2\mu)X_{12} + \lambda X_{13}$$

$$Y_{13} = \lambda X_{11} + \lambda X_{12} + (\lambda+2\mu)X_{13}$$

$$Y_{14} = \mu X_{14}$$

$$Y_{22} = \lambda X_{21} + (\lambda+2\mu)X_{22} + \lambda X_{23}$$

$$Y_{23} = \lambda X_{21} + \lambda X_{22} + (\lambda + 2\mu) X_{23}$$

$$Y_{24} = \mu X_{24}$$

$$Y_{33} = \lambda X_{31} + \lambda X_{32} + (\lambda + 2\mu) X_{33}; Y_{34} = \mu X_{34}$$

$$Y_{44} = \mu X_{44} \quad \dots (7)$$

and $X_{11} = [(\lambda + 2\mu)r_1^2 + \lambda r_1 r_2 + \lambda r_1 r_3]$

$$X_{12} = [(\lambda + 2\mu)r_1 r_2 + \lambda r_2^2 + \lambda r_2 r_3]$$

$$X_{13} = [(\lambda + 2\mu)r_1 r_3 + \lambda r_2 r_3 + \lambda r_3^2]$$

$$X_{14} = [(\lambda + 2\mu)r_1 r_4 + \lambda r_2 r_4 + \lambda r_3 r_4]$$

$$X_{21} = [\lambda r_1^2 + (\lambda + 2\mu)r_1 r_2 + \lambda r_1 r_3]$$

$$X_{22} = [\lambda r_1 r_2 + (\lambda + 2\mu)r_2^2 + \lambda r_2 r_3]$$

$$X_{23} = [\lambda r_1 r_3 + (\lambda + 2\mu)r_2 r_3 + \lambda r_3^2]$$

$$X_{24} = [\lambda r_1 r_4 + (\lambda + 2\mu)r_2 r_4 + \lambda r_3 r_4]$$

$$X_{31} = [\lambda r_1^2 + \lambda r_2 r_1 + (\lambda + 2\mu)r_3 r_1]$$

$$X_{32} = [\lambda r_1 r_2 + \lambda r_2^2 + (\lambda + 2\mu)r_3 r_2]$$

$$X_{33} = [\lambda r_1 r_3 + \lambda r_2 r_3 + (\lambda + 2\mu) r_3^2]$$

$$X_{34} = [\lambda r_1 r_4 + \lambda r_2 r_4 + (\lambda + 2\mu) r_3 r_4]$$

$$X_{41} = \mu r_4 r_1$$

$$X_{42} = \mu r_4 r_2$$

$$X_{43} = \mu r_4 r_3$$

$$X_{44} = \mu r_4^2 \quad \dots (8)$$

Typical term in Eqn. (6) is expressed as follows:

$$\begin{aligned} Y_{11} &= (\lambda + 2\mu) [(\lambda + 2\mu) r_1^2 + \lambda r_1 r_2 + \lambda r_1 r_3] + \lambda [(\lambda + 2\mu) r_1 r_2 \\ &\quad + \lambda r_2^2 + \lambda r_2 r_3] + \lambda [(\lambda + 2\mu) r_1 r_3 + \lambda r_2 r_3 + \lambda r_3^2] \\ &= (\lambda + 2\mu)^2 r_1^2 + 2\lambda(\lambda + 2\mu) (r_1 r_2 + r_3 r_1) + \lambda^2 r_2^2 + \lambda^2 r_2 r_3 \\ &\quad + \lambda^2 r_2 r_3 + \lambda^2 r_3^2 \\ &= [(\lambda + 2\mu) r_1 + \lambda r_2 + \lambda r_3]^2 \\ &= [(\lambda + 2\mu) \left(a + \frac{S_x}{2\sqrt{J_2}}\right) + \lambda \left(a + \frac{S_y}{2\sqrt{J_2}}\right) + \lambda \left(a + \frac{S_z}{2\sqrt{J_2}}\right)]^2 \\ &= \left[3\lambda a + \frac{\lambda}{2\sqrt{J_2}} (S_x + S_y + S_z) + 2\mu a + \frac{2\mu}{2\sqrt{J_2}} (S_x)\right]^2 \end{aligned}$$

$$\begin{aligned}
&= [3\lambda a + 2\mu a + \frac{\mu}{\sqrt{J_2}} S_x]^2 \\
&= [a(3\lambda + 2\mu) + \frac{\mu}{\sqrt{J_2}} S_x]^2 = E_{11}^2 \quad \dots (9)
\end{aligned}$$

$$\begin{aligned}
Y_{22} &= \lambda X_{21} + (\lambda + 2\mu) X_{22} + \lambda X_{23} \\
&= \lambda [\lambda r_1^2 + (\lambda + 2\mu) r_1 r_2 + \lambda r_1 r_3] + (\lambda + 2\mu) [\lambda r_1 r_2 \\
&\quad + (\lambda + 2\mu) r_2^2 + \lambda r_2 r_3] + \lambda [\lambda r_1 r_3 + (\lambda + 2\mu) r_2 r_3 + \lambda r_3^2] \\
&= [(\lambda + 2\mu) r_2 + \lambda r_1 + \lambda r_3]^2 \\
&= [(\lambda + 2\mu) (a + \frac{S_y}{2\sqrt{J_2}}) + \lambda (a + \frac{S_x}{2\sqrt{J_2}}) + \lambda (a + \frac{S_z}{2\sqrt{J_2}})]^2 \\
&= [a(3\lambda + 2\mu) + \frac{\mu}{\sqrt{J_2}} S_y]^2 = E_{22}^2 \quad \dots (10)
\end{aligned}$$

$$\begin{aligned}
Y_{14} &= [(\lambda + 2\mu) r_1 r_4 + \lambda r_2 r_4 + \lambda r_3 r_4] \\
&= \mu\lambda (r_1 r_4 + r_2 r_4 + r_3 r_4) + 2\mu^2 r_1 r_4 \\
&= \{ \mu\lambda [(a + \frac{S_x}{2\sqrt{J_2}}) + (a + \frac{S_y}{2\sqrt{J_2}}) + (a + \frac{S_z}{2\sqrt{J_2}}) \\
&\quad + 2\mu^2 (a + \frac{S_x}{2\sqrt{J_2}})] \frac{\tau_{xy}}{\sqrt{J_2}} \\
&= [3\mu\lambda a + 2\mu^2 a + \mu^2 \frac{S_x}{\sqrt{J_2}}] \frac{\tau_{xy}}{\sqrt{J_2}}
\end{aligned}$$

$$\begin{aligned}
&= [a\mu(3\lambda + 2\mu) + \mu^2 \frac{S_x}{\sqrt{J_2}}] \frac{\tau_{xy}}{\sqrt{J_2}} \\
&= [a(3\lambda + 2\mu) + \mu \frac{S_x}{\sqrt{J_2}}] \frac{\mu\tau_{xy}}{\sqrt{J_2}} = E_{11} E_{12} \quad \dots (11)
\end{aligned}$$

$$\begin{aligned}
Y_{24} &= \mu X_{24} \\
&= \mu [\lambda r_1 r_4 + (\lambda + 2\mu) r_2 r_4 + \lambda r_3 r_4] \\
&= [\lambda\mu(r_1 + r_2 + r_3) + 2\mu r_2] r_4 \\
&= [a(3\lambda + 2\mu) + \mu \frac{S_y}{\sqrt{J_2}}] \frac{\mu\tau_{xy}}{\sqrt{J_2}} = E_{22} E_{12} \quad \dots (12)
\end{aligned}$$

$$\begin{aligned}
Y_{33} &= \lambda X_{31} + \lambda X_{32} + (\lambda + 2\mu) X_{33} \\
&= \lambda [\lambda r_1^2 + \lambda r_2 r_1 + (\lambda + 2\mu) r_1 r_3] \\
&\quad + \lambda [\lambda r_1 r_2 + \lambda r_2^2 + (\lambda + 2\mu) r_3 r_2] \\
&\quad + (\lambda + 2\mu) [\lambda r_1 r_3 + \lambda r_2 r_3 + (\lambda + 2\mu) r_3^2] \\
&= 2\lambda(\lambda + 2\mu) [r_2 r_3 + r_1 r_3] + (\lambda + 2\mu)^2 r_3^2 + \lambda^2 r_2^2 + \lambda^2 r_1^2 \\
&\quad + 2\lambda^2 [r_1 r_2] \\
&= [(\lambda + 2\mu) r_3 + \lambda r_2 + \lambda r_1]^2
\end{aligned}$$

$$\begin{aligned}
&= \left[(\lambda+2\mu) \left(a + \frac{S_z}{2\sqrt{J_2}} \right) + \lambda \left(a + \frac{S_y}{2\sqrt{J_2}} \right) + \lambda \left(a + \frac{S_x}{2\sqrt{J_2}} \right) \right]^2 \\
&= \left[a(3\lambda+2\mu) + \frac{\mu}{\sqrt{J_2}} S_z \right]^2 = E_{33}^2 \quad \dots (13)
\end{aligned}$$

$$\begin{aligned}
Y_{34} &= \mu X_{34} \\
&= \mu [\lambda r_1 r_4 + \lambda r_2 r_4 + (\lambda+2\mu) r_3 r_4] \\
&= [\lambda \mu (r_1 + r_2 + r_3) + 2\mu^2 r_3] r_4 \\
&= \left[a(3\lambda+2\mu) + \mu \frac{S_z}{\sqrt{J_2}} \right] \frac{\mu \tau_{xy}}{\sqrt{J_2}} = E_{33} E_{12} \quad \dots (14)
\end{aligned}$$

$$\begin{aligned}
Y_{44} &= \mu X_{44} \\
&= \mu [\mu r_4^2] \\
&= (\mu r_4)^2 = \left(\mu \frac{\tau_{xy}}{\sqrt{J_2}} \right)^2 = E_{12}^2 \quad \dots (15)
\end{aligned}$$

Similarly the terms Y_{12} , Y_{13} and Y_{23} are derived as follows:

$$\begin{aligned}
Y_{12} &= \lambda X_{11} + (\lambda+2\mu) X_{12} + \lambda X_{13} \\
&= \lambda^2 (r_1^2 + r_2^2 + r_3^2) + 2\lambda(\lambda+\mu) (r_1 r_2 + r_2 r_3 + r_1 r_3)
\end{aligned}$$

$$\begin{aligned}
& + 2\mu\lambda(r_1^2 + r_2^2) + 4\mu^2 r_1 r_2 + 2\mu\lambda r_1 r_2 \\
= & [\lambda(r_1+r_2+r_3)]^2 + 2\mu\lambda[3a^2 + \frac{1}{4J_2} (S_x S_y + S_y S_z + S_z S_x)] \\
& + 2\mu\lambda(r_1 + r_2)^2 + 4\mu^2 r_1 r_2 + 2\mu\lambda r_1 r_2
\end{aligned}$$

Putting $S_z = -(S_x+S_y)$ one obtains

$$\begin{aligned}
Y_{12} = & [\lambda(r_1+r_2+r_3)]^2 + 2\mu\lambda[3a^2 + \frac{1}{4J_2} (-S_x^2 - S_y^2 - S_x S_y)] \\
& + 2\mu\lambda [a^2 + \frac{S_x^2}{4J_2} + \frac{aS_x}{\sqrt{J_2}} + a^2 + \frac{S_y^2}{4J_2} + \frac{aS_y}{\sqrt{J_2}}] \\
& + 4\mu^2 r_1 r_2 + 2\mu\lambda r_1 r_2 \\
= & (3a\lambda)^2 + 2\mu\lambda(3a^2) + 2\mu\lambda[3a^2 + \frac{3a(S_x+S_y)}{2\sqrt{J_2}}] \\
& + 4\mu^2 r_1 r_2 \\
= & (3a\lambda)^2 + 2\mu\lambda[6a^2 + \frac{3a(S_x+S_y)}{2\sqrt{J_2}}] + 4\mu^2 (a + \frac{S_x}{2\sqrt{J_2}}) (a + \frac{S_y}{2\sqrt{J_2}}) \\
= & a^2 [3\lambda+2\mu]^2 + \frac{\mu a}{\sqrt{J_2}} (3\lambda+2\mu) (S_x+S_y) + \mu^2 \frac{S_x S_y}{J_2} \\
= & [a(3\lambda+2\mu) + \frac{\mu S_x}{\sqrt{J_2}}] [a(3\lambda+2\mu) + \frac{\mu S_y}{\sqrt{J_2}}] = E_{11} E_{12} \dots (16)
\end{aligned}$$

Following the derivation for Y_{12} , it can be shown

$$\begin{aligned}
 Y_{13} &= \lambda X_{11} + \lambda X_{12} + (\lambda + 2\mu) X_{13} \\
 &= [a(3\lambda + 2\mu) + \frac{\mu S_x}{\sqrt{J_2}}] [a(3\lambda + 2\mu) + \frac{\mu S_z}{\sqrt{J_2}}] = E_{11} E_{33} \dots (17)
 \end{aligned}$$

and finally

$$\begin{aligned}
 Y_{23} &= \lambda X_{21} + \lambda X_{22} + (\lambda + 2\mu) X_{23} \\
 &= [a(3\lambda + 2\mu) + \frac{\mu S_y}{\sqrt{J_2}}] [a(3\lambda + 2\mu) + \frac{\mu S_z}{\sqrt{J_2}}] = E_{22} E_{33} \dots (18)
 \end{aligned}$$

APPENDIX - C
DESCRIPTION OF COMPUTER PROGRAM' OPFA'

The element library at present consists of a four noded isoparametric quadrilateral element (Zienkiewicz, 1971) specified in the plane of analysis ($r - z$, or $y - z$ plane). The computer programme has features viz dynamic storage, automatic core size adjustment, element and nodal point generation, an equation block solver and a variety of output options.

The following steps describe in general the procedure normally adopted in a two-dimensional soil-structure interaction analysis with the programme OPFA:

- Step 1 Read general information such as title, number of elements, number of nodal points etc;
- Step 2 Read input data such as nodal point and element data, material properties for constitutive modelling and boundary conditions;
- Step 3 Read input data for the forcing function (time history of loading) and transfer these data to create global load vector in time domain;
- Step 4 Form total mass matrix;
- Step 5 Form total stiffness matrix for either undrained or drained condition;
- Step 6 Form total damping matrix which includes

the usual damping matrix for the solid part and dissipation matrix for the fluid flow;

Step 7 Solve the system of equations of motion, linear or non-linear in the time domain.

Based on the input information as provided for the constitutive modelling of the soil beneath the foundation, the programme OPFA branches to different subsections.

For a linear elastic analysis, the global stiffness matrix for static or dynamic analysis is formed only once and the information regarding responses are passed to Step 8.

For equivalent linear analysis, the effective shear strains are computed for each element and based on these shear strain amplitudes, strain-compatible soil properties such as shear moduli and damping ratios are determined. These values are compared with the initial properties as used in the analysis and steps 4, 5, 6, 7 are repeated with new properties if required. Once the desired convergence is achieved, response information such as displacements are passed to Step 8.

For an incremental nonlinear analysis, the global stiffness matrix is updated at each time step based on the stress-state of each element. If any element has yielded while using a particular failure model, an elastic-plastic constitutive matrix is assigned to that element. However, elements which have not yielded, are assigned the usual elastic constitutive matrices. During each time or load step increment, an iterative approach in combination with the well known modified Newton-Raphson (MNR) technique is used with the total applied load subdivided into a finite number of load increments. For a dynamic analysis this means the evaluation of the effective load vector at each time step.

For each load/time increment, the incremental deflection is determined using the modified Newton-Raphson technique. If the solution at the start of a load increment is known, the tangent stiffness matrix at that point can be computed and used to obtain an initial linear estimate of the deflection at the end of the increment. However, since the load-deflection or stress-strain curve of the

material is nonlinear, the actual load the system can support at this deflection is less than the applied load and thus a load unbalance exists at the end of the first iteration. The unbalanced loads are applied to the system and a revised estimate of the incremental deflection is determined using the same tangent stiffness matrix evaluated at the previous point. Once the desired convergence is achieved, response information such as displacements are passed to Step 8.

In the shakedown analysis an elastic analysis is first carried out for all possible load combinations and initial elastic limit loads are determined for various loading combinations. Next the global compatibility matrix, and constitutive matrix are formed in order to solve the mathematical optimization problem. The result obtained provides the shakedown load factor.

Step 8 Compute the final stress response from the converged displacement using stress-displacement matrix.

The detailed description regarding the internal structure of the programme OPFA and an user's guide with detailed comments on input data are given in the following sections of the Appendix.

The programme OPFA is written in FORTRAN IV language and was originally developed on the IBM-370/158 computer at the Memorial University of Newfoundland, St. John's, Canada.

C.2 OPERATIONAL MODES

Basically the programme OPFA can be used in five different modes of operation. These modes are controlled by a parameter (variable) called 'NDYN'. Depending on the value 'NDYN' as input by the user, the particular type of analysis mode will be selected for the entire problem. These modes are described below:

- a) NDYN. EQ.1 - Dynamic analysis with time history of loading function only (e.g. wave loading or any other type of loading)
- b) NDYN. EQ.3 - Calculation of eigenvalues and eigenvectors only ie. eigenvalue analysis
- c) NDYN. EQ.4 - Static analysis only
- d) NDYN. EQ.5 - Transient phenomenon only e.g. consolidation problem

The material modelling is primarily controlled by a parameter called 'KKS'. The following sets of value for 'KKS' can be used in order to handle a specific type of material model. Table C1 presents the various combinations of material modellings (e.g. elastic, equivalent linear, elastic-plastic and piecewise linear) that can be used with different modes of analyses.

- a) KKS. EQ.1 - Linear elastic analysis (constant soil properties)
- b) KKS. EQ.2 - Equivalent linear soil model with strain compatible soil properties (e.g. variable stiffness and damping)
- c) KKS. EQ.3 - Nonlinear soil model with incremental elastic-plastic analysis
- d) KKS. EQ.4 - Shakedown analysis with piecewise linearized yield surface for soil; linear programming technique

C.3 SUBPROGRAMMES

The programme OPFA consists of a main programme and 61 subroutines. The flow chart of OPFA is shown in Fig. C1, and the calling sequences of the different subroutines are given in Table C2. This table also describes which subroutines control the 'READ' and/or 'WRITE' operation of the twenty-one (21) logical tape devices used as temporary storage files and the two physical tapes (or mass storage files) used for input and output of data. The main programme and 9 other major subroutines control the basic sequences of operations.

TABLE C1 - MODES OF OPERATIONS IN PROGRAM OPFA

MODE OF OPERATION	DESCRIPTION OF OPERATION	CONSTITUTIVE MODELLING COMBINATION	REMARKS
Mode 1 NDYN = 1	Time history analysis with superstructure loading	<ul style="list-style-type: none"> - Linear (KKS = 1) - Equivalent linear (KKS = 2) - Incremental analysis (KKS = 3) with nonlinear soil model - Shakedown analysis with piecewise linearized yield surface for soil (KKS = 4) 	
Mode 3 NDYN = 3	Frequency analysis; Eigenvalues and Eigenvectors	<ul style="list-style-type: none"> - Linear soil model (KKS = 1) - Equivalent linear Soil model (KKS = 2) 	
Mode 4 NDYN = 4	Static or quasi - static response analysis only	<ul style="list-style-type: none"> - Linear (KKS = 1) - Incremental analysis (KKS = 3) with nonlinear soil model - Shakedown analysis with piecewise linearized yield surface for soil (KKS = 4) 	
Mode 5 NDYN = 5	Transient static response analysis only e.g. Consolidation problem	<ul style="list-style-type: none"> - Linear (KKS = 1) - Incremental analysis with nonlinear soil (KKS = 3) model 	

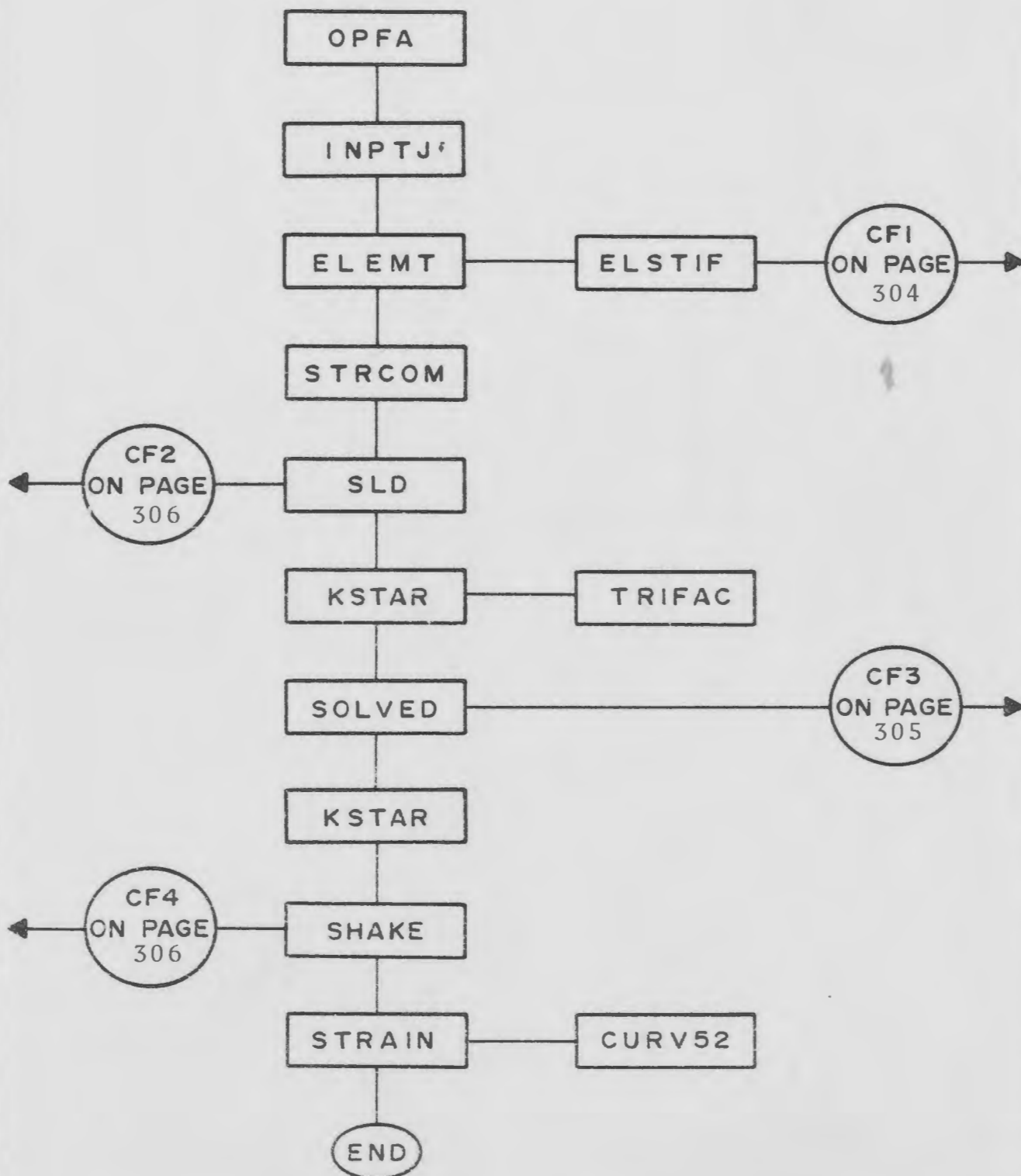
FIG. C1 — FLOW DIAGRAM FOR OPFA.

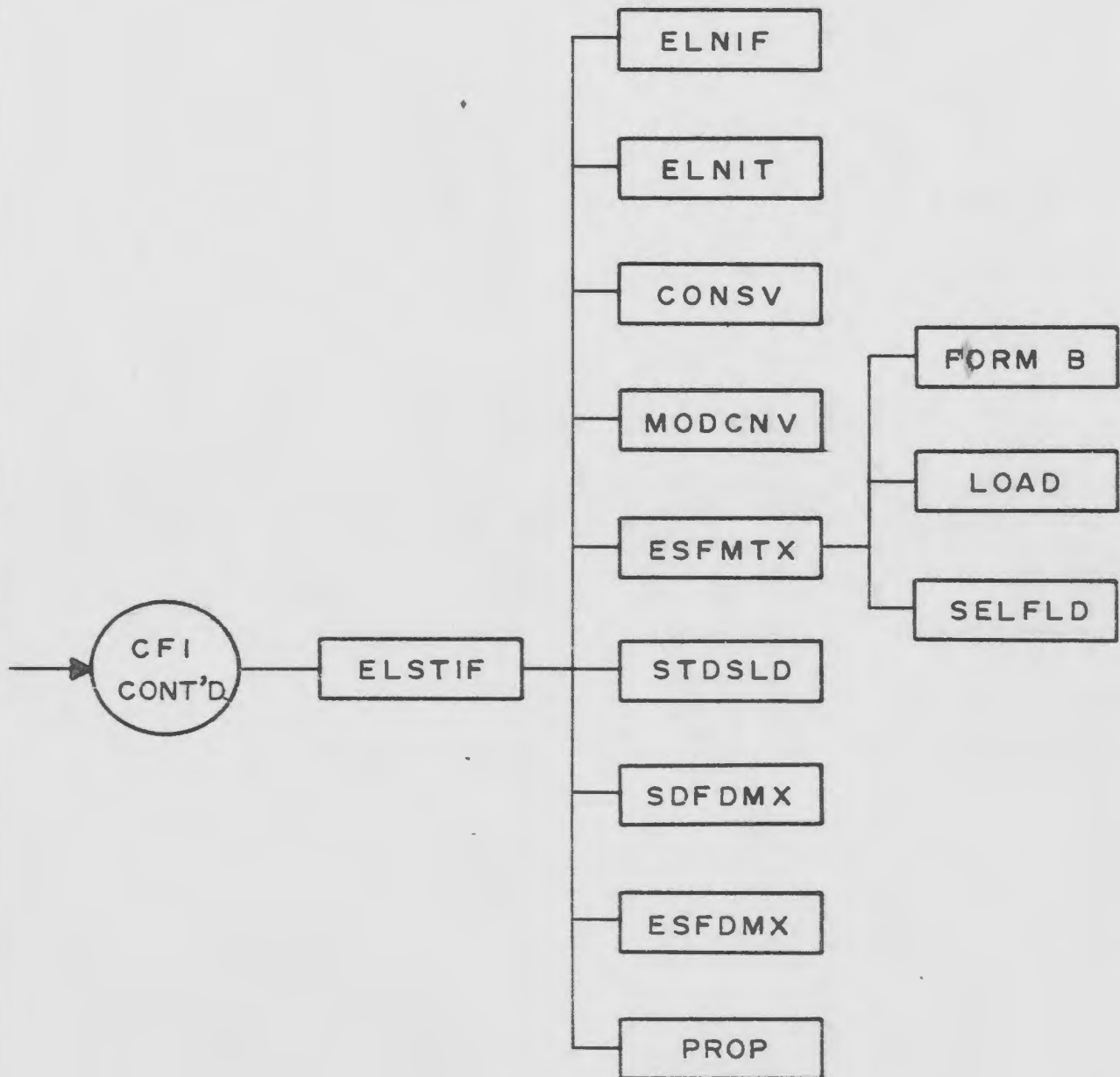
FIG. C1 — FLOW DIAGRAM FOR OPFA. (CONT'D.)

FIG. C1 — FLOW DIAGRAM FOR OPFA. (CONT'D.)

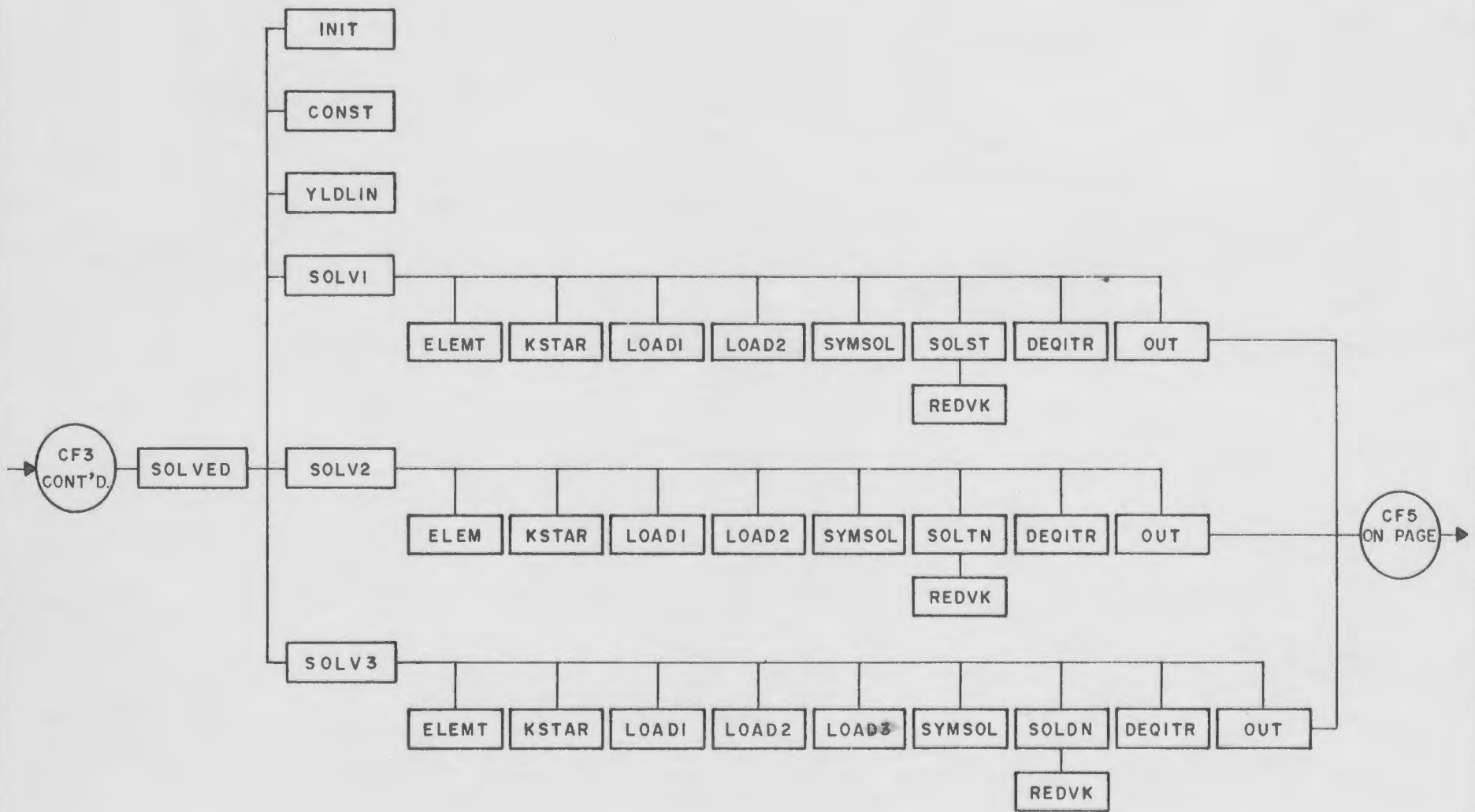


FIG. C1 — FLOW DIAGRAM FOR OPFA. (CONT'D.)

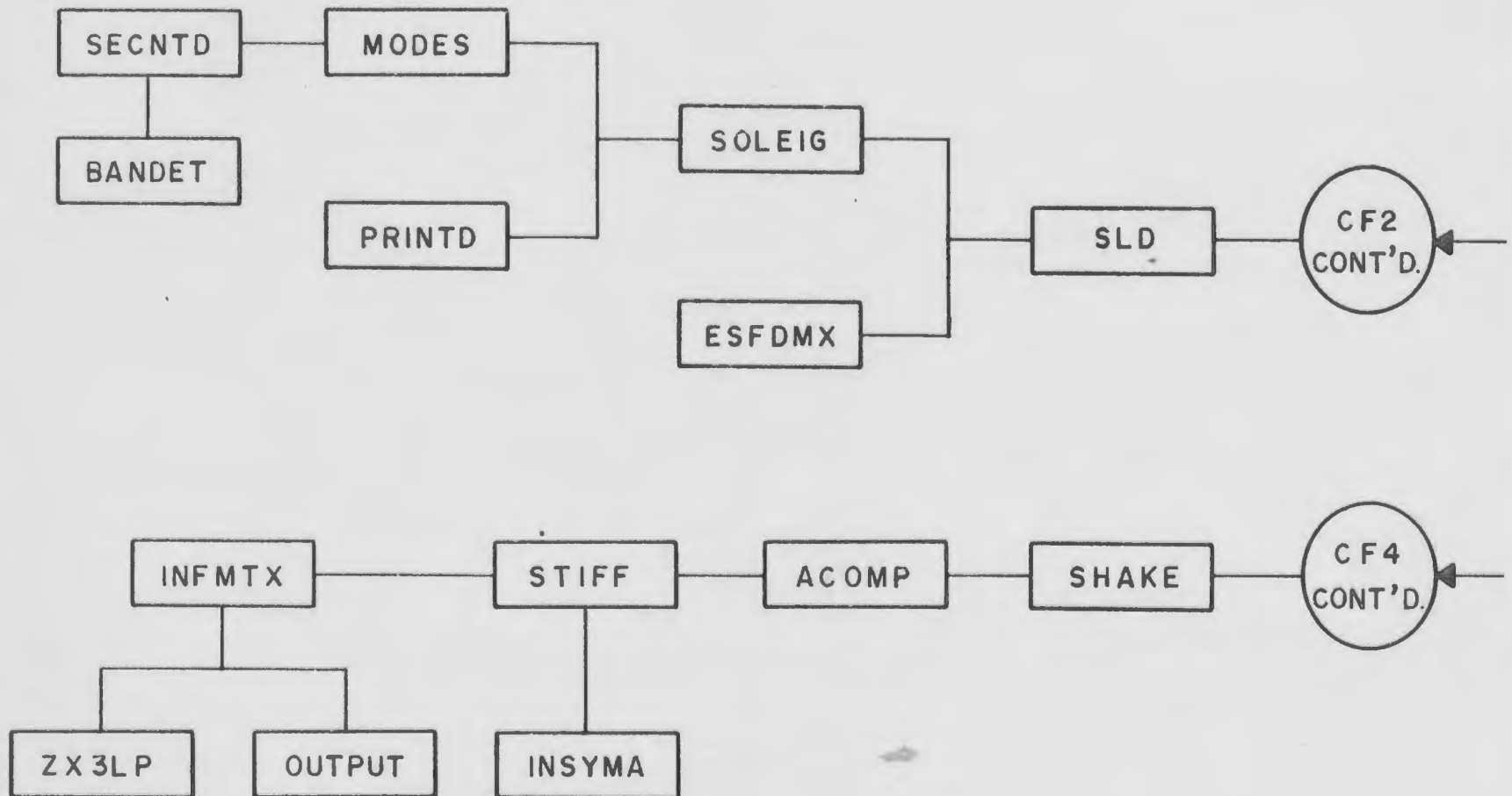


FIG. C1 — FLOW DIAGRAM FOR OPFA. (CONT'D.)

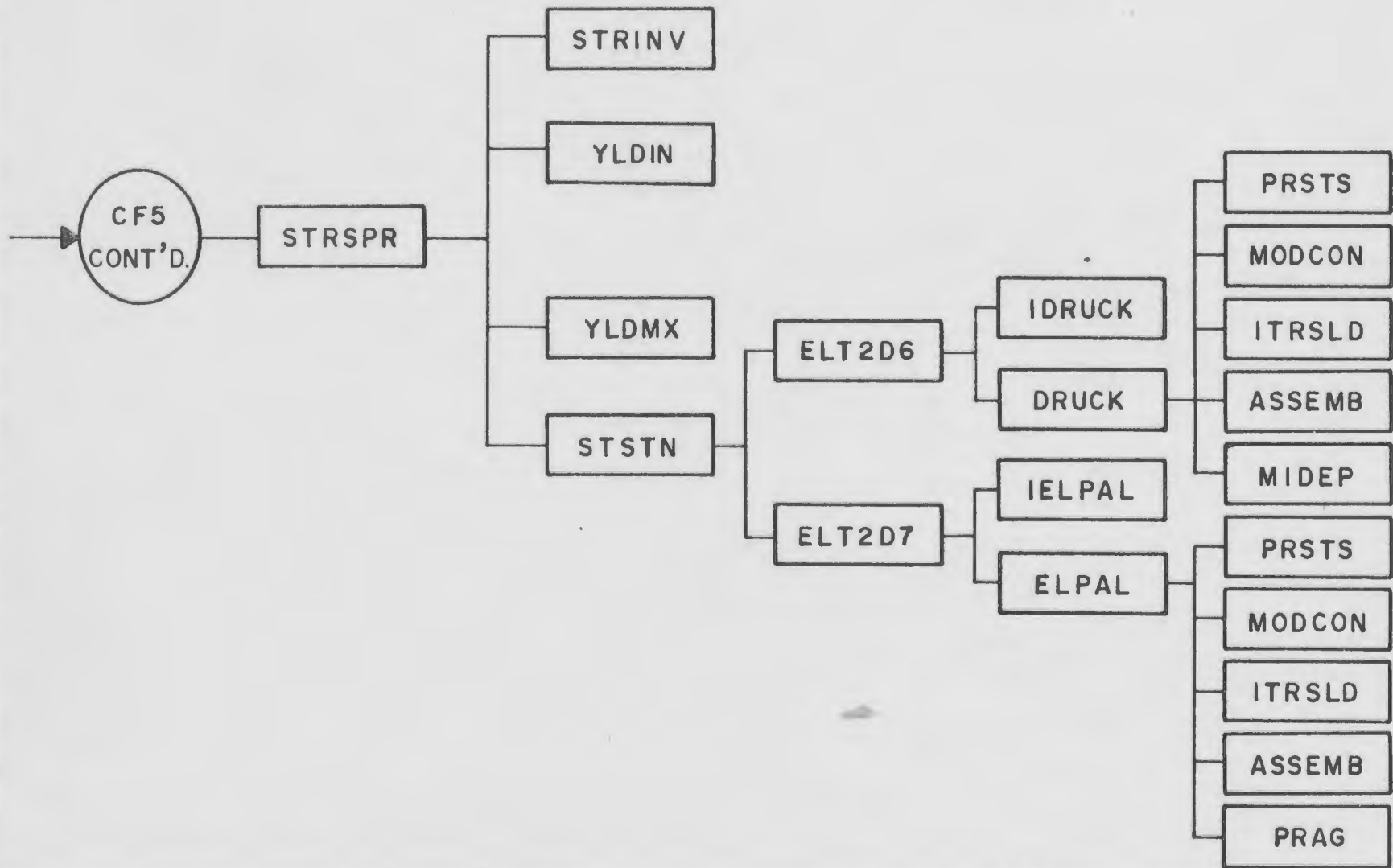


TABLE C2 - CALLING SEQUENCE FOR SUBROUTINES
AND TAPES IN PROGRAM OPFA

NAME OF ROUTINE	ROUTINE CALLS	ROUTINE CALLED BY	ROUTINE WRITES ON TAPE	ROUTINE READS FROM TAPE
ELT2D6	IELPAL & ELPAL	STSTN		
ELT2D7	IDRUCK & DRUCK	STSTN		
ELSTIF	ELNIF, ELINT, ELINT, ELINT, CONSV, MODCNV, ESFMTX, STDSL, STDSL, SDFDMX, ESFDMX, PROP	ELEMT	11, 13	
ELNIF		ELSTIF	14	14
ELINT		ELSTIF		
ESFDMX		ELSTIF		
ESFMTX	FORMB, LOAD MODCNV, SELF LD	ELSTIF		
FORMB		ESFMTX		
EQCHK		SOLV1, SOLV2, & SOLV3		60
IDRUCK		ELT2D7		
IELPAL		ELT2D6		
INFMTX	ZX3LP & OUTPUT	SHAKE		

TABLE C2 - CALLING SEQUENCE FOR SUBROUTINES
AND TAPES IN PROGRAM OPFA (CONT'D)

NAME OF ROUTINE	ROUTINE CALLS	ROUTINE CALLED BY	ROUTINE WRITES ON TAPE	ROUTINE READS FROM TAPE
INIT		SOLVED		
INPUTJ		MAIN		
INSYMA		STIFF		
ITRSLD		ELPAL, DRUCK STRSPR		
KSTAR	TRIFAC		4,9,60	13
LOAD		ESFMTX		
LOAD1		SOLV1, SOLV2 & SOLV3		
LOAD2		SOLV1, SOLV2 & SOLV3		
MAIN	(See Section A.3.1)			
MODCNV		ELSTIF		
MODES	SECNTD & BANDET	SOLEIG		10
MIDEP		ELPAL		
PRAGER		DRUCK		

TABLE C2 - CALLING SEQUENCE FOR SUBROUTINES
AND TAPES IN PROGRAM OPFA (CONT'D)

NAME OF ROUTINE	ROUTINE CALLS	ROUTINE CALLED BY	ROUTINE WRITES ON TAPE	ROUTINE READS FROM TAPE
PRINTD		SOLEIG		8
PROP		ELSTIF		
PRSTS		ELPAL & DRUCK		
OUT	ASSEMB	SOLV1, SOLV2 & SOLV3	55	55, 57, 18
OUTPUT	CHECK	INFMTX		
REDVK		SOLST, SOLTN, DEQITR, SOLDN		3
MULT		BANDET		
MAXMIN		ELPAL, DRUCK		
MDCNEQ		STRSPR	17	
SDF DMX		ELSTIF	51	
SECNTD	BANDET	MODES	8	9, 4,
SELF LD		ESFMTX		
SHAKE	ACOMP, STIFF, INFMTX	MAIN		

TABLE C2 - CALLING SEQUENCE FOR SUBROUTINES
AND TAPES IN PROGRAM OPFA (CONT'D)

NAME OF ROUTINE	ROUTINE CALLS	ROUTINE CALLED BY	ROUTINE WRITES ON TAPE	ROUTINE READS FROM TAPE
SLD	SOLEIG & ESFDMX	MAIN	61, 4, 9, 13	51, 22, 61,
SOLDN	REDVK	SOLV3		
SOLEIG	MODES & PRINTD	SLD	14	14
SOLST	REDVK	SOLV1		
SOLTN	REDVK	SOLV2		
SOLVED	INIT, CONST, YLDLIN, SOLV1, SOLV2 & SOLV3	MAIN		
SOLV1	ELEMT, KSTAR, LOAD1, LOAD2, SYMSOL, SOLST, DEQITR, OUT	SOLVED		
SOLV2	ELEMT, KSTAR, LOAD1, LOAD2, SYMSOL, SOLTN, DEQITR, OUT	SOLVED		
SOLV3	ELEMT, KSTAR, LOAD1, LOAD2, SYMSOL, SOLDN, DEQITR, OUT	SOLVED	12	12
STDSL	VECTOR	ELSTIF	51	
STIFF	INSYMA	SHAKE		

TABLE C2 - CALLING SEQUENCE FOR SUBROUTINES
AND TAPES IN PROGRAM OPFA (CONT'D)

NAME OF ROUTINE	ROUTINE CALLS	ROUTINE CALLED BY	ROUTINE WRITES ON TAPE	ROUTINE READS FROM TAPE
STRAIN	CURV52	MAIN	22	51,22
STRSPR	YLDMX, STSTN, ITRSLD	OUT	18, 57	10, 11, 17, 22, 50
STSTN	ELT2D6 & ELT2D7	STRSPR		
SYMSOL		SOLV1, SOLV2 & SOLV3		
TRIFAC		KSTAR	3	4, 1, 8,
YLDIN		STRSPR		
YLDLIN		SOLVED		
YLDCHK		MAIN		11
YLDMX		STRSPR		
STRINV		STRSPR		
ZX3LP		INFMTX		
VECTOR		STDSL		

C.3.1 PROGRAM CONTROL

The principal subroutines are detailed in this section and a brief description of the remaining subroutines are given in Table C3.

The programme OPFA calls the main subroutines INPUTJ, ELEMT, STRCOM, SLD, KSTAR, SOLVED, YLDCHK, SHAKE and STRAIN.

The subroutine INPUTJ reads or generates the input data for nodal point coordinates and boundary conditions. For each node the maximum number of allowable displacement boundary conditions that can be prescribed is four i.e. two degrees of freedom (D.O.F.) for solid and two degrees of freedom (D.O.F.) for fluid with respect to solid. These boundary conditions are created by means of a two dimensional array called the nodal connectivity array or ID-array. Based on the active degrees of freedom for each node, the total number of equations for the entire problem is created in this routine. This routine returns with the information on nodal point coordinates, degrees of freedom array (ID-array) and the total number of equations.

The subroutine ELEMT via ELSTIF reads the element

information such as nodal connection, material constitutive properties such as bulk modulus, shear modulus, porosity, permeability, fluid bulk modulus, densities of bulk solid and fluid, damping information etc. Based on these properties, this subroutine computes and stores on low speed tapes the following sets of information for each element.

- a) element mass matrix (only lumped masses)
- b) element stiffness matrix including solid, fluid and the solid-fluid coupling submatrices
- c) element damping matrix including the dissipation matrix defined as the function of the permeability of the soil medium
- d) element stress-displacement matrix

The subroutine STRCOM reads all material properties information related to the equivalent linear soil model for a dynamic finite element analysis. The equivalent linear soil model is primarily based on strain compatible soil properties and this routine reads these properties such as maximum shear moduli, maximum damping ratios, initial shear moduli, poisson's ratios etc.

The subroutine SLD formulates the solid mass and stiffness matrices either for an equivalent linear analysis or for a shakedown analysis assuming an

undrained or drained condition. In the equivalent linear soil model, the fundamental frequency is computed based on the global mass and stiffness matrices. Using this frequency, Rayleigh damping coefficients are evaluated for each element and thus damping matrices for the entire finite element model can be evaluated. For shakedown analysis, this subroutine returns the global assembled stiffness matrix for the two-phase medium either in the undrained or/drained condition as specified.

The subroutine KSTAR assembles all matrices such as mass, stiffness, damping and load vector computed for each element by imposing the prescribed boundary condition. These matrices are read from a low speed storage tape. The solution is normally sought by in-core method and as such these are stored on a low speed storage tape. This routine also returns the decomposed effective assembled stiffness matrix $[\hat{K}]$ using a standard tridiagonal factorization (LDL^T) method. In the incremental nonlinear analysis, the stiffness matrix needs to be updated at every time or load step increment and therefore this routine will be called and updated 'NCYCL' number of times where 'NCYCL' represents the total number of time or load increment steps. However, for other types of analyses, such as elastic, equivalent linear or

shakedown, the global stiffness matrix is constant over the entire time duration of the loading period and therefore needs no updating.

The subroutine SOLVED initializes all variables necessary for the solution of the equations of motion. It also sets up all constants required for the step-by-step integration scheme.

For linear analysis ie. 'KKS = 1', the equations of motion are solved and the stresses and strains are computed from the nodal displacement vector either for a single time step (NDYN = 4) or for the entire time history (NDYN = 1).

For equivalent linear analysis ie. 'KKS = 2', the equations of motion are solved in the time domain and the strain history for each element is determined from the nodal displacement vector.

For incremental nonlinear analysis, ie. 'KKS = 3', the equations of motion are solved in an iterative manner and element strains and stresses are determined from the converged nodal displacement vector obtained from the last iteration.

For shakedown analysis i.e. 'KKS = 4', the linear

elastic analysis is performed first in order to determine the initial elastic limit load. This is done for all possible combination of loadings. The global compatibility matrix is then formed in order to solve the mathematical optimization problem.

C.4 TAPES

The programme OPFA uses a total of 21 tapes, as shown in Table C4. The two physical tapes, #5 and #6 are used to read input data from, and/or write output data on magnetic tapes or mass storage files. The twenty-one (21) logical tape devices i.e. tapes 1 to 4, 8 to 14, 17, 18, 21, 22, 50, 51, 55, 57, 60 and 61 are used for low speed storage of temporary data during the execution mode of the programme. The content of these tapes is described in Table C4. As described previously, Table C2 indicates which subroutines control the 'READ' and/or 'WRITE' operations of all tapes used with the exception of tapes 5, 6 and 7. These three tapes are input and output files which represent a card reader, on-line printer and on-line card punch, respectively.

TABLE C3 - BRIEF DESCRIPTION OF SUBROUTINES
IN PROGRAM OPFA

NAME OF ROUTINE	DESCRIPTION OF OPERATION
ACOMP	computes global assembled compatibility and constitutive matrices for the shakedown analysis.
ASSEMB	assembles and forms the global residual load vector for incremental nonlinear analysis.
BANDET	extracts the first eigenvalue of the system.
CHECK	computes the global residual force vector for the system.
COMPGT	computes the strain-displacement relationship at each gauss point.
CONST	computes the constants for numerical time step integration.
CONSV	computes the elasticity constants of the constitutive matrix for each element under drained and/or undrained condition.
CURV52	computes the reduction factor for shear modulus and damping ratio for each element based on the maximum shear strain evaluated from the strain history.
DEQITR	performs the equilibrium iteration at each load or time step.
DRUCK	computes incremental strain and stress; checks the DRUCKER-PRAGER yield condition and forms the elastic or elastic-plastic constitutive matrix depending on the stress state of the particular element at the current load and/or time step.

TABLE C3 - BRIEF DESCRIPTION OF SUBROUTINES
IN PROGRAM OPFA (CONT'D)

NAME OF ROUTINE	DESCRIPTION OF OPERATION
ELEMT	for description see Section C.3.
ELPAL	computes incremental strain and stress; checks the Von Mises yield condition and forms at each time and/or load step, the elastic or elastic-plastic constitutive matrix for each element depending on the stress state of the particular element.
ELINT	initializes all variables related to the formulation of the element matrices.
ELNIF	reads all necessary information related to each element, eg. element nodal connectivity, material properties, etc.
ELSTIF	computes element mass matrices (solid and fluid), stiffness matrices (solid, fluid and solid-fluid coupling) and damping matrices (solid and fluid).
ELT2D6	organizes the storage locations for all variables related to the elastic-plastic analysis based on Von Mises yield condition.
ELT2D7	organizes the storage locations for all variables related to the elastic-plastic analysis based on Drucker-Prager yield condition.
EQCHK	checks the overall equilibrium at each load/time step.
ESFDMX	Forms element matrices with respect to solid-fluid coupling.

TABLE C3 - BRIEF DESCRIPTION OF SUBROUTINES
IN PROGRAM OPFA (CONT'D)

NAME OF ROUTINE	DESCRIPTION OF OPERATION
ESFMTX	computes element submatrices such as mass, damping and stiffness with respect to solid and fluid part.
IDRUCK	initializes all variables before assigning them to 'DRUCK'.
IELPAL	initialize all variables before assigning them to 'ELPAL'.
INFMTX	sets up the 'tableau' based on dual problem and finds the shakedown load factor by calling a standard linear programming routine (ZX3LP).
INIT	initializes all variables before assigning them to the relevant routines for the solution of the equations of motion.
INPUTJ	reads all nodal point data such as node number, degrees of freedom and coordinates.
INSYMA	inverts a full matrix of size (NEQ X NEQ).
ITRSLD	computes the nodal residual load on each element based on the stresses evaluated at each gauss point.
KSTAR	for description see Section C.3.1.
LOAD	computes residual load vector (nodal) for each element based on the stress states at each gauss point. This routine is required for elastic-plastic analysis.

TABLE C3 - BRIEF DESCRIPTION OF SUBROUTINES
IN PROGRAMME OPFA (CONT'D)

NAME OF ROUTINE	DESCRIPTION OF OPERATION
LOAD1	reads time histories of surface loading and transfers these loadings to global force vector in time domain.
LOAD2	reads time histories of concentrated loads and transfers these loadings to global force vector in time domain.
MAIN	for description see Section C.3.1.
MDCNEQ	computes the elasticity constants for equivalent linear analysis.
MODCNV	reallocate the elastic-plastic constitutive matrix for subsequent iteration.
MODCON	reallocate the stress vector for elastic-plastic analysis.
MODES	organizes the storage allocations for the eigenvalue analysis.
MIDEP	computes the incremental stress-strain relationship using the Von Mises yield condition.
PRAGER	computes the incremental stress-strain relationship using the Drucker-Prager yield condition.
PRINTD	prints the eigenvalues and eigenvectors.
PROP	reads the material properties such as hardening, yield strength, cohesion, and angle of internal friction.

TABLE C3 - BRIEF DESCRIPTION OF SUBROUTINES
IN PROGRAMME OPFA (CONT'D)

NAME OF ROUTINE	DESCRIPTION OF OPERATION
PRSTS	stores the stress computed at each gauss point in an array form.
OUT	prints all displacements, velocities, acceleration and also creates a punch file to save these data for subsequent plotting.
OUTPUT	computes permanent inelastic strains and prints them.
REDVK	performs the solution of the equilibrium equations after reading the decomposed stiffness matrix from the tape.
MULT	performs the multiplication of a matrix by a vector.
SDFDMX	computes the element undrained stiffness matrix and the associated solid-fluid coupling matrix.
SECNTD	computes the eigenvalues and eigenvectors of a banded matrix by the determinant search solution.
SHAKE	for description see Section C.3.1.
SLD	for description see Section C.3.1.
SOLDN	solves the dynamic equilibrium equations of motion at a particular load and/or time step.

TABLE C3 - BRIEF DESCRIPTION OF SUBROUTINES
IN PROGRAMME OPFA (CONT'D)

NAME OF ROUTINE	DESCRIPTION OF OPERATION
SOLEIG	reads the data required for an eigenvalue analysis.
SOLST	solves the static equilibrium equations at a particular time and/or load step.
SOLTN	solves the transient static equilibrium equations at a particular time and/or load step.
SOLVED	for description see Section C.3.1.
SOLV1	solves the static problem.
SOLV2	solves the time dependent quasi-static problem.
SOLV3	solves the dynamic problem.
STDSL D	computes the stress-displacement matrix and condenses the element stiffness matrix for incompatible nodes.
STIFF	converts the banded stiffness matrix to a full matrix before assigning it to 'INSYMA'.
STRAIN	for description see Section C.3.1.
STRCOM	for description see Section C.3.1.
STRSPR	computes element stress, strains from the nodal displacements.

TABLE C3 - BRIEF DESCRIPTION OF SUBROUTINES
IN PROGRAMME OPFA (CONT'D)

NAME OF ROUTINE	DESCRIPTION OF OPERATION
STSTN	calls two key routines 'ELT2D6' and 'ELT2D7'.
SYMSOL	solves a set of simultaneous equations.
TRIFAC	decomposes the banded effective stiffness matrix.
YLDIN	initializes the variables before assigning them to routine 'YLDMX'.
YLDLIN	computes gradients of the piecewise linearized yield planes for each material type and writes the coefficients of the gradient matrices on a tape.
YLDCHK	computes the elastic limit load for each load case.
YLDMX	computes the maximum stress responses of each element normalized with respect to each yield plane.
STRINV	computes the normal strain components from stresses for each element.
VECTOR	computes the vector of a matrix.
ZX3LP	solves the linear programming problem using the revised simplex algorithm.

TABLE C4 - DESCRIPTION OF TAPES
IN PROGRAMME 'OPFA'

TAPE NUMBER	DESCRIPTION OF CONTENT
1	auxiliary tapes used at the time of eigenvalue analysis or at the time of the solution of equations of motion.
2	contains all control information regarding eigenvalue solution.
3	contains all information regarding the factorized decomposed global stiffness matrix for the entire problem.
4	contains the assembled global stiffness matrix in banded form.
8	contains all information regarding eigenvectors in the event an eigenvalue analysis is required; However, the same file is also used in the decomposition of the stiffness matrix for the solution of system of linear or nonlinear equations of motion. In the later case, the tape is primarily used as an auxiliary one.
9	contains the assembled global mass matrix.
10	contains information on given starting iteration vectors for an eigenvalue analysis; However, the same tape unit is also used to store the coefficients of the gradients of the piecewise linearized yield planes if a shakedown analysis is requested.
11	contains information on element stress-displacement matrices for solid

TABLE C4 - DESCRIPTION OF TAPES
IN PROGRAMME 'OPFA' (CONT'D.)

TAPE NUMBER	DESCRIPTION OF CONTENT
	and fluid parts, element connectivity array, elastic material properties, gauss point coordinates, and volume related to each gauss point.
12	contains information on load vectors in the event an equivalent linear analysis is carried out. In the first iteration the load vectors are formed once and written on this tape for use in subsequent iteration.
13	contains information on element matrices related to connectivity arrays, solid-fluid coupling matrices including solid and fluid stiffness matrices, solid and fluid mass matrices, damping matrices respectively.
14	contains all input data regarding element types, element nodal numbering, material types, element material properties such as Lamé's constants, poisson's ratio, etc.
17	contains matrices of coefficients of elastic constants for all elements under drained condition.
18	contains the information on stresses for selected element groups for contour plotting.
21	contains the nodal load vector computed for each element based on the stress state at each gauss point.
22	contains information on maximum shear

TABLE C4 - DESCRIPTION OF TAPES
IN PROGRAMME 'OPFA' (CONT'D.)

TAPE NUMBER	DESCRIPTION OF CONTENT
	moduli, maximum damping ratios, and shear moduli values used for all elements. This tape is required for an equivalent linear soil model with the dynamic analysis option.
50	contains matrices of coefficients of elastic constants for all elements under the undrained condition. This tape is required in the event a shakedown analysis is performed.
51	contains element connectivity arrays, element stiffness matrices (solid and fluid) and mass matrices (solid and fluid); it also contains element undrained stiffness matrices, fluid stiffness matrices and coupling solid-fluid matrices, etc.
55	contains information on displacement, velocities and acceleration for selected nodal points at selected time steps. These data are later used for plotting purposes.
57	contain information on stresses, and strains on selected elements at specified time-steps. These data are later processed for plotting purposes.
60	contains global stiffness matrix for equilibrium check.
61	contains information regarding element matrices and properties, volume, densities and solid, fluid stiffness matrices including coupling terms; The tape is required in the event KKS = 2 ie. equivalent linear soil model.

C.5 COMMENTS ON PROGRAMME CAPACITY DYNAMIC STORAGE AND
RUN TIME

The programme OPFA has been developed on an IBM 370 computer using FORTRAN IV language. During the execution mode of any problem, the core requirement is determined as a function of the following variable as described below:

- NUMNP - Total number of nodal points;
- NUMEL - Total number of elements;
- NUMMAT - Total number of material types;
- NCON - Number of parameters required to describe a particular soil model for incremental analysis.
- IDWA - Number defining the storage requirements for each element stress-strain information together with the yield condition. (whether elastic or plastic). For KKS = 3, IDWA will vary from 10 to 11 depending on the material model;
- NEQ - Number of equations in the finite element model;
- MBAND - Half-band width;
- NF - Number of frequencies to be evaluated;

NACL	-	1;
LL	-	Number of maximum surface load cards used for all load cases;
NNYP	-	Number of elements * number of yield planes;
NNSC	-	Number of elements * number of stress components;
NLC	-	Number of load cases;
NYP	-	Number of yield planes;
NSC	-	Number of stress components.

The length of the dynamic storage array depends on the amount of data to be stored at different stages of the execution and hence will vary during a run. In order to account for this variation and to minimize the dynamic storage space required, the programme has been developed with a special feature. This feature allows the expansion or contractions of the core size required in accordance with the length of the blank common array, n , during the different execution steps.

C.5.1 High Speed Storage Requirements:

The high speed storage requirements of the programme can be changed depending on the size of the problem to be solved. This is done by changing the two

FORTTRAN statements at the beginning of the OPFA programme ie.

```
COMMON D(n)
```

```
MTOT = n
```

The minimum value of n needed is computed as follows except for the shakedown analysis which is controlled by the parameter KKS. EQ.4 as described previously. For the shakedown analysis using the linear programming technique, OPFA requires a larger size of storage allocation even for a very small/moderate size problem. However, for all other types of analysis e.g. linear, equivalent linear and incremental nonlinear analysis, the storage requirement is controlled by the few internal parameters e.g. $N1$, $N5$, $NK15$, $N011$, $NN15$, $NN17$ and $N43$. The value of $N1$ is set as 1 at the beginning of the programme and the value of 'n' is determined from the parameter $N43$ ie. setting n equal to $N43$. The following key equations describe the determination of the internal parameters as listed above:

$$N1 = 1$$

$$N5 = N1 + 8 * NUMNP$$

$$NK15 = N5 + 11 * NUMMAT + NUMEL + NCON * \\ NUMMAT + IDWA * NUMEL$$

$$N011 = NK15 + 3 * NEQ * MBAND + 3 * NEQ + 2 \\ * NEQ * MBAND$$

$$\text{NN15} = \text{N011} + (3 * \text{NEQ} + \text{MBAND} - 1) + 9 * \text{NUMMAT}$$

$$\begin{aligned} \text{NN17} = & \text{NN15} + 1 + \text{NEQ} * \text{MAXO}(\text{MBAND}, \text{NF} + 3) \\ & + 16 * \text{NEQ} + 4 * (\text{NF} + 3) + 3 * \text{NF} + \\ & 2 * \text{NUMNP} \end{aligned}$$

$$\begin{aligned} \text{N43} = & (\text{NN17} \text{ or } \text{NN15}) + 3 * \text{NEQ} + 8 * \text{NUMNP} \\ & (\text{or } \text{NEQ}) + 12 * \text{NEQ} + 2 * \text{NACL} + 18 * \\ & \text{NUMEL} + 10 * \text{NUMNP} + 4 * 2 * \text{LL} * 2 + \\ & 2 * \text{NNYP} + \text{NNSC} * \text{NNYP} + \text{NLC} + \text{NYP} + \\ & \text{NYP} * \text{NYP} \end{aligned}$$

If the value of 'n' is set less than the required value as determined by N43, an error message is printed and programme execution is terminated indicating the particular value of the internal parameter and the name of the particular subroutine at termination.

APPENDIX - D
INPUT DATA TO 'OPFA'

I. HEADING CARD (10A8)

Notes	Columns	Variable	Description of Input
(1)	1-80	HED(8)	Read heading information to be printed with the output

NOTES/

- (1) Begin each new data case with a new heading card.

II. MASTER CONTROL CARD (9I5)

Notes	Columns	Variable	Description of Input
(1)	1-5	NUMNP	Total number of nodal points
(2)	6-10	NUMEL	Total number of elements
(3)	11-15	NDYN	Analysis type code; EQ.1; Forced dynamic response analysis EQ.3; Eigenvalues/vector solutions EQ.4; Static analysis EQ.5; Quasi-static transient analysis
(4)	16-20	IPLNAX	EQ.-1; Plane stress model EQ.0; Plane strain model

II. MASTER CONTROL CARD - CONT'D

Notes	Columns	Variable	Description of Input
(5)	21-25	LL	GE.1; Maximum number of surface load cards for the entire problem
(6)	26-30	NF	Number of frequencies to be found in the eigenvalue analysis EQ.0; Static analysis GE.1; Dynamic analysis
(7)	31-35	MODEX	EQ.0; No data check
(8)	36-40	NAD	EQ.0; Superstructure not included in the analysis EQ.1; Superstructure included in the analysis
(9)	41-45	NGAMA	EQ.0; Body force not included

NOTES/

- (1) Total number of nodal points; Nodal points are numbered sequentially from '1' to 'NUMNP'. For comments on nodal points data generation see section VIII.
- (2) Total number of elements; Elements are numbered sequentially from '1' to 'NUMEL'. For comments on element data generation see section XI.

II. MASTER CONTROL CARD - CONT'D

NOTES/ - CONT'D

- (3) If NDYN. EQ.1 the programme performs a forced dynamic response analysis.
- If NDYN. EQ.3 the programme performs an eigenvalue analysis only and returns with the eigenvalues and eigenvectors.
- If NDYN. EQ.4 the programme performs a static analysis only.
- (4) The variable 'IPLNAX' controls the type of two dimensional modelling which has to be selected for a particular problem.
- (5) At least one (1) surface load card (LL.GE.1) is required whether or not the user inputs the surface pressure loading. For details see section XXII and XXIII.
- (6) In the event of an equivalent linear analysis using 'KKS' = 2, (see next section for reference), NF should be greater than or equal to 1. This is required only when a dynamic equivalent linear analysis with strain compatible soil properties is carried out. However, for NDYN. EQ.3 (eigenvalue analysis only), NF shall also be greater than or equal to 1.
- (8) If NAD. EQ.0, no superstructure/foundation is included in the analysis i.e. load is applied directly to the soil. However, by setting NAD.NE.0 a superstructure such as a concrete structure may be included in the analysis.
- (9) 'NGAMA' EQ.0 represents soil overburden pressure i.e. body force has not been included in the analysis.

III OUTPUT PRINT & PUNCH CONTROL CARD (3I5)

Notes	Columns	Variable	Description of Input
(1)	1-5	ISTRPR	GE.1; Stress output interval
(2)	6-10	NPRINT	GE.1; Displacement, velocity and acceleration output interval
(3)	11-15	ISTART	GE.1; Stress/strain output interval for stress/strain contour plotting

NOTES/

- (1) Every ISTRPR-th time point responses such as stresses, strains at each element will be printed and also saved for plotting purpose.
- (2) Every NPRINT-th time point responses such as displacement velocity and acceleration for each nodal point will be printed and saved for subsequent plotting.
- (3) Every ISTART-th time point, output data for stress/strain will be saved for contour plotting.

IV MASTER CONTROL CARD (I5, F10.0, I5, F10.0, 5I5)

Notes	Columns	Variable	Description of Input
(1)	1-5	NCYCL	GE.1; Total number of time steps or load steps
(2)	6-15	DT	GT.0; Solution time step for numerical integration,

IV MASTER CONTROL CARD - CONT'D

Notes	Columns	Variable	Description of Input
(3)	16-20	KKS	EQ.1; Elastic analysis EQ.2; Equivalent linear analysis with strain compatible soil properties (variable stiffness and damping) EQ.3; Incremental non-linear analysis with elastic-plastic soil model EQ.4; Shakedown analysis with piecewise linearized yield surface
(4)	21-30	FA	Gravitational constant (value of 'g')
(5)	31-35	NDRN	EQ.0; Undrained condition EQ.1; Undrained - drained condition (flow problem)
(6)	36-40	NCON	NCON = 2; Required only if KKS = 3. Otherwise set NCON = 1
(7)	41-45	IDW	IDW = 10 or 11 depending on the parameter 'Model' as described below. Required only if KKS = 3; otherwise set IDW = 1

IV MASTER CONTROL CARD - CONT'D

Notes	Columns	Variable	Description of Input
(8)	46-50	MODEL	EQ.1; Von Mises (IDW = 10) EQ.2; Drucker-Prager (IDW = 11) Required only when KKS = 3; otherwise leave blank
(9)	51-55	NINT	Number of integration points; set NINT = 2

NOTES/

- (1) 'NCYCL' controls the total number of time steps.
- (2) The value of 'DT' defines the time step increment. The programme has been written based on constant time steps. However, this may sometimes be disadvantageous when an analysis, 'NDYN. EQ.5 (say, consolidation problem), is carried out because of the large numbers of 'NCYCL' steps.
- (3) 'KKS' is one of the key parameters which controls the type of analysis to be performed based on a particular material model as selected.

If KKS. EQ.1 a linear elastic analysis will be performed ie. material properties will remain entirely constant during the analysis.

If KKS. EQ.2, an equivalent linear analysis following the procedure as described by Seed and Idriss (1969) will be carried out. This procedure takes into account the variable damping and shear modulus for each element depending on its strain level.

IV MASTER CONTROL CARD - CONT'D

NOTES/ - CONT'D

If KKS. EQ.3, an incremental nonlinear (elastic-plastic) analysis is performed based on either one of the yield models as selected. The two types of yield models which are at present available are: 1) Von Mises yield criteria with or without isotropic hardening and 2) Drucker-Prager yield criteria with zero hardening (elastic - perfectly plastic).

If KKS. EQ.4, a shakedown analysis is performed using piecewise linearized yield model ie. modelling the quadratic yield function with a suitable number of yield planes. The linear programming technique is then used to evaluate the shakedown load factor.

- (4) Input the gravitational constant value for computation of the mass matrix;
- (5) The parameter 'NDRN' controls the state of the soil masses under the loading conditions.

If NDRN. EQ.0, this will represent a purely undrained situation i.e. no movement of fluid with respect to solid and as such all degrees of freedom related to fluid are fixed.

VV SELECTIVE INPUT DATA (4E10.3)

Notes	Columns	Variable	Description of Input
(1)	1-10	AREA	GT.0.0; Area of the foundation (Equivalent model of a circular foundation)

V SELECTIVE INPUT DATA (4E10.3) - CONT'D

Notes	Columns	Variable	Description of Input
(2)	11-20	WIDTH	GT.0.0; Width of the footing used in plane strain modelling
(3)	21-20	WEIGHT	GT.0.0; Submerged weight of the platform
(4)	31-40	SHEAR	GT.0.0; Undrained shear strength of the soil beneath the foundation

NOTES/

- (1) In this section the equivalent area of the foundation is input.
- (2) Width of the equivalent plane strain footing is input here.
- (3) Submerged or buoyant weight of the platform is input here.
- (4) Undrained soil shear strength beneath the foundation; As soil shear strength will vary over the depth, the shear strength for the top layer is normally input here.

VI A CONTROL CARD FOR DISPLACEMENT OUTPUT (5X, 15)

Notes	Columns	Variable	Description of Input
(1)	5-10	*JPT	GE.0; Number of nodal points for which plotting is required

VIII NODAL POINT DATA (515, 2F10.0, 15)

Notes	Columns	Variable	Description of Input
(2)	6-10	ID (N,1)	X-translation boundary condition code for solid
	10-15	ID (N,2)	Y-translation boundary condition code for solid
	16-20	ID (N,3)	X-translation boundary condition code for fluid
	21-25	ID (N,4)	Y-translation boundary condition code for fluid
(3)	26-35	R (N)	X - coordinate
(4)	36-45	Y (N)	Y - coordinate
(5)	46-50	KM	Node number increment

NOTES/

(1) Nodal point data must be defined for all (NUMNP) nodes. Nodal point data may be input directly (ie. each node on its own individual card) or the generatin option may be used if applicable (see note 5, below). Nodal point numbering should range from "1" to the total number of nodes 'NUMNP' sequentially.

(2) Boundary condition codes can only have the following sets of assigned values describing the movement of the solid and fluid.

(M = 1, 2, 3 and 4) ID(N,M) = 0; unspecified (free) displacement component

VIII NODAL POINT DATA - CONT'D

NOTES/ - CONT'D

ID(N,M) = 1/-1; Deleted
displacement
component

- (3) & (4) Two dimensional coordinate of a node; X or Y
in cartesian system.
- (5) Nodal point generation parameter; always given on
the second card of a sequence.

IX MATERIAL CONTROL CARD (3I5)

Notes	Columns	Variable	Description of Input
(1)	1-5	NUMMAT	GE.1; Number of material property set
(2)	6-10	NELYTP	EQ.0; 4-noded isopara- metric element EQ.1; 4-noded isopara- metric element with incompatible modes
(3)	11-15	NS	Stress output option; EQ.0; Centroid and gauss points EQ.20; Midside

X MATERIAL PROPERTIES CARD (8E10.3)

For each material identification, two cards must be
supplied in order to define each unique set of properties.

X MATERIAL PROPERTIES CARD (8E10.3) - CONT'D

Notes	Columns	Variable	Description of Input
	1-10	ELMDA(N)	Lame's constant - λ (drained)
	11-20	EMD(N)	Lames's constant - μ (drained)
	21-30	EALFA(N)	Solid grain compress- ibility - α
	31-40	EM(N)	Water bulk modulus - K_w
	41-50	EK(N)	Permeability - k
	51-60	DENB(N)	Bulk density - ρ
	61-70	DENF(N)	Water density - ρ_f
	71-80	FK(N)	Porosity - n

Carry over to next card;

	1-10	DBT1(N)	Rayleigh damping coefficients - B_1
	11-20	DBT2(N)	- do - B_2

NOTES/

XI ELEMENT CARDS (5I3, 5I2, 5E10.3, F5.2)

Notes	Columns	Variable	Description of Input
(1)	1-3	INEL	Element number
(2)	4-6	I	Number of nodal point I
	7-9	J	Number of nodal point J
	10-12	K	Number of nodal point K
	13-15	L	Number of nodal point L
(3)	16-17	IMAT	GE.1; Material identification number; (i.e. NUMMAT)
(4)	18-19	KN	GE.0; Element generation code;
(5)	20-21	KJUMP	EQ.0;
(6)	22-23	IKS	EQ.0; Free movement of solid
			EQ.1; Fixed movement
(7)	24-25	IKF	EQ.0; Free movement of fluid
			EQ.1; Fixed movement
(9)	76-80	THICK	GE.0; Element thickness

XII MATERIAL PROPERTY CARD - ELASTIC-
PLASTIC ANALYSIS (I5, 2E12.5)

*** THIS INPUT IS REQUIRED IN THE EVENT KKS = 3;
OTHERWISE SKIP THE FOLLOWING SECTION. ****

Notes	Columns	Variable	Description of Input
(1)	1-5	N	LE. NUMMAT; Material identifica- tion number
(2)	6-18	YIELD(N)/ COH(N)	For MODEL = 1; Read 'YIELD(N)' For MODEL = 2; Read 'COH(N)'
(3)	19-31	HARD(N)/ PHI(N)	For MODEL = 1; Read 'HARD(N)' as hardening parameter For MODEL = 2; Read 'PHI(N)' as angle of internal friction

NOTES/

- (2) & (3) Incremental elastic-plastic analysis can be carried out based on two types of material modelling. These are defined as; MODEL = 1; Von Mises yield criteria and MODEL = 2; Drucker-Prager yield criteria. Von Mises material model requires the yield strength and hardening modulus as input parameters. Hardening modulus, E_p , can be expressed as some percentage of the original linear elastic modulus. The actual value can be determined from an uniaxial stress/strain curve. However, for the Drucker-Prager type material model, the analysis can be carried out assuming the material behaves as an elastic-perfectly plastic manner. In this case, the value of E_p will be zero (no hardening) and the only material properties to be input are $c' = \text{Cohesion}$ and $\phi_u = \text{angle of internal friction}$.

XIII CONTROL CARD FOR NUMERICAL INTEGRATION

Notes	Columns	Variable	Description of Input
(1)	1-10	TETA	For static analysis = 1.0; However, for dynamic analysis $\theta = 1.0$; constant acceleration method or $\theta = 1.4$; WILSON - θ method
	11-20	BETA	= 0.0;
	21-30	GAMA	= 0.0;
(2)	31-40	RTOL	Tolerance value; Required if KKS = 3; otherwise leave blank
(3)	41-45	NPUNCH	EQ.0; Output do not require saving
			EQ.1; Output to be saved for subsequent plotting
(4)	46-50	NEQUIB	Normally set equal to zero;
			However, for KKS = 3; NEQUIB can be set equal to 1
(5)	51-55	ITEMX	Number of iterations specified within each time or load step;

XIII CONTROL CARD FOR NUMERICAL INTEGRATION - CONT'D

NOTES/

- (1) For static analysis ' θ ' is normally chosen as 1.0, ie. equal time step or load step is followed. However, for dynamic analysis ' θ ' could be anywhere between 1.0 to 1.4. For $\theta = 1.0$, this represents constant acceleration method whereas ' $\theta = 1.4$ ', represents 'WILSON- θ ' method. In the 'WILSON - θ ' method load vector and stiffness matrices are constructed at an extended time step ie. at, $t = t + \tau$, where $\tau = \theta \Delta t$, and equilibrium equations are solved at ' $t + \tau$ ' time step. The displacement, velocity, and acceleration at time ' $t + \Delta t$ ' are then computed from the corresponding values at time ' $t + \theta \Delta t$ '. For all dynamic analysis, time step ' Δt ' as defined in sectin IV should be sufficiently small in order to get a stable, meaningful result.
- (2) Tolerance value required to carry out the incremental elastic-plastic analysis (KKS = 3). If NEQUIB. EQ.0, RTOL is set equal to 0.0. However, when NEQUIB. EQ.1, RTOL is set equal to some small percentage of the load increment. Care and judgement should be exercised in order to avoid poor convergence of the results. The load increment or time increment should be also sufficiently small in order to avoid the divergence of the residual load distribution with respect to the actual incremental load that has been applied on the system.

XIV CONTROL CARD FOR LOAD CASES, LINEARIZED YIELD PLANES AND ITERATION NUMBER

Notes	Columns	Variable	Description of Input
(1)	1-5	NLC	GE.1; Number of load cases
(2)	6-10	NP3	EQ.3; Number of stress components. Required only when KKS = 4; otherwise set as '1'

XIV CONTROL CARD FOR LOAD CASES, LINEARIZED
YIELD PLANES AND ITERATION NUMBER - CONT'D

Notes	Columns	Variable	Description of Input
(3)	11-15	NP6	Number of yield planes; varies from 6 to 14. Plane strain model NP6 = 6 whereas plane stress model NP6 = 14; Required only when KKS = 4; otherwise set equal to '1'
(4)	16-20	NUMBER	Total number of iterations required in order to perform an equivalent linear analysis and to obtain strain compatible dynamic soil properties; Required in the event KKS = 2; otherwise set "NUMBER" as 1

XV MATERIAL PROPERTY CARD FOR EQ. LINEAR
ANALYSIS (5X, 4E12.5)

*** IF (KKS. NE.2) SKIP THIS SECTION ***

One card is required for each unique set of properties.

Notes	Columns	Variable	Description of Input
(1)	6-18	GMX(N)	Maximum shear modulus evaluated at small strain
(2)	19-31	DPL(N)	Maximum critical damping ratio at small strain
(3)	32-43	GUSE(N)	Shear modulus used in the first iteration

XV MATERIAL PROPERTY CARD FOR EQ. LINEAR ANALYSIS (5X, 4E12.5) - CONT'D

Notes	Columns	Variable	Description of Input
(4)	44-56	PO(N)	Poisson's ratio

NOTES/

- (1) Shear modulus at $10^{-4}\%$ strain, see Chapter 5.
- (2) Critical damping ratio at small strain (refer Chapter 5 and Table 5.1).
- (3) Value of shear modulus used in the first iteration.

XVI CONTROL CARD TO IDENTIFY FOOTING/STRUCTURE ELEMENTS

** IF (KKS. NE.2) SKIP THIS SECTION**

Notes	Columns	Variable	Description of Input
(1)	1-5	NOEL	GE.0; Number of footing elements
(2)	6-80	(NL(I), I = 1,NOEL)	Elements numbered sequentially

NOTES/

- (1) Total number of footing or structural elements in the model
- (2) Element numbers entered sequentially in order to model the foundation/structure.

XVII CONTROL CARD FOR EIGEN VALUE ANALYSIS
(3I5, 2F10.0, I5)

** IF (KKS EQ.2 or NDYN. EQ.3). THIS SECTION IS REQUIRED TO BE INPUT, OTHERWISE SKIP **

Notes	Columns	Variable	Description of Input
(1)	1-5	IFPR	Flag for printing intermediate matrices, norms, etc; calculated during the eigenvalue solution EQ.0; Do not print EQ.1; Print
(2)	6-10	IFSS	Flag for performing the STRUM SEQUENCE check EQ.0; Check to see if eigenvalues were missed EQ.1; Pass on the check
(3)	11-15	NITEM	Maximum number of iteration allowed to reach the convergence tolerance; EQ.0; default set to '16'
(4)	16-25	RTOL	Convergence tolerance (accuracy) for the highest (NF) request eigenvalue;

XVII CONTROL CARD FOR EIGEN VALUE ANALYSIS
(3I5, 2F10.0, I5) - CONT'D

Notes	Columns	Variable	Description of Input
(5)	26-35	COFQ	Cut off frequency (cycles/unit time); EQ.0; NF eigenvalues will be extracted GT.0; Extract only those values below COFQ
(6)	36-40	NFO	Number of starting iteration vectors to be read from tape 10

XVIII CONTROL CARD FOR SHAKEDOWN ANALYSIS (4I5)

** IF (KKS. NE.4) SKIP THIS SECTION **

Notes	Columns	Variable	Description of Input
(1)	1-5	IPT	EQ.1; Minimize the objective function
(2)	16-20	NPRINT	EQ.0; No intermediate print is required EQ.1; Intermediate prints are required

XIX CONTROL CARD FOR MATERIAL PROPERTY; MATERIAL
TYPE FOR LINEARIZER MODEL, (I5)

** If one card is required for each set of unique
property**

XIX CONTROL CARD FOR MATERIAL PROPERTY; MATERIAL
TYPE FOR LINEARIZER MODEL, (I5) - CONT'D

Notes	Columns	Variable	Description of Input
(1)	1-5	ITYPE(N)	EQ.1; None for KKS = 1, 2 & 3 EQ.0; For KKS = 4 and IPLNAX = 0; Plane strain model

NOTES/

- (1) 'ITYPE(N)' is a parameter which controls the type of piecewise linearized yield model to be used in the shakedown analysis.

For plane stress condition ie. IPLNAX = -1, only Von Mises model is available. This model is described by fourteen yield planes.

However, for plane strain condition ie. IPLNAX = 0, three types of piecewise linearized yield model are available at present. These are a) Von Mises, b) Tresca and c) Mohr-Coulomb.

XX CONTROL CARD - MATERIAL PROPERTIES (2E10.3)

Notes	Columns	Variable	Description of Input
(1)	1-10	COH(N)	Cohesion value
(2)	11-20	PHI(N)	Angle of internal friction value

XXI CONTROL CARD FOR SURFACE LOAD DATA

Notes	Columns	Variable	Description of Input
(1)	1-5	ISLC	EQ.0; No concentrated loading is prescribed NE.0; Required concentrated loading as input
(2)	6-80	(NSLCI(K), K = 1, NLC)	Number of surface load cards NSLCI(K) for each 'K' load case. (where K. GE.1 but LE.NLC)

NOTES/

- (1) Input data which governs whether a concentrated loading has to be prescribed or not.
- (2) Number of surface loading data cards for each load case; 'NSLC' = NSLC(K), K = 1, NLC, where 'NSLC' represents number of surface loading cards for a particular load case and 'NLC' indicates total number of load cases.

XXII CONTROL CARD FOR SCALE FACTORS WITH SURFACE LOADING-BULK (8E10.3)

Notes	Columns	Variable	Description of Input
(1)	1-80	(SCALB(I), I = 1, NSLC)	Scale factor for each surface loading card; NSLC represents the number of surface loading cards for one load case.

XXII CONTROL CARD FOR SCALE FACTORS WITH
SURFACE LOADING-BULK (8E10.3) - CONT'D

NOTES/

- (1) The surface pressures for bulk loading (as input in section XXV on page 358) are amplified by the scale factor 'SCALB(I)' for each surface loading data card.

XXIII CONTROL CARD FOR SCALE FACTORS WITH
SURFACE LOADING - FLUID (8E10.3)

** IF (NDRN. EQ.0) SKIP THIS SECTION BELOW. **

Notes	Columns	Variable	Description of Input
(1)	1-8	(SCALB(I), I = 1, NSLC)	Description of input scale factor for fluid loading for each surface loading card; 'NSLC' represents the number of surface loading cards for one load case

NOTES/

- (1) The surface intensity pressures for fluid loading as input in section XXV on page 358 are amplified by this scale factor 'SCALB(I)' for each surface loading data card.

XXIV CONTROL CARD FOR SURFACE LOADING DATA INPUT

A) TIME POINTS (8F10.0)

Notes	Columns	Variable	Description of Input
(1)	1-10	ID(N)	First time point of the data entry; Read the data as a pair of two data sets; TD(1), & TD(2), etc.

B) SURFACE LOAD DATA (2I4, 4E11.4) - BULK LOAD

Notes	Columns	Variable	Description of Input
(1)	1-4	ISC(L)	Node indicating 'I'
(2)	5-8	JSC(L)	Node indicating 'J'
(3)	9-19	SURTRX (N,L,1)	X-direction pressure intensity on node 'I'
(4)	20-30	SURTRX (N,L,2)	X-direction pressure intensity on node 'J'
(5)	31-41	SURTRY (N,L,1)	Y-direction pressure intensity on node 'I'
(6)	42-52	SURTRY (N,L,2)	Y-direction pressure intensity on node 'J'

C) SURFACE LOAD DATA FOR FLUID PRESSURE (2I4, 4E11.4)

** IF (NDRN. EQ.0) SKIP THIS SECTION BELOW. **

Notes	Columns	Variable	Description of Input
	1-4	ISC(L)	Node indicating 'I'

XXIV CONTROL CARD FOR SURFACE LOADING DATA INPUT - CONT'D

C) SURFACE LOAD DATA FOR FULLD PRESSURE - CONT'D

Notes	Columns	Variable	Description of Input
	5-8	JSC(L)	Node indicating 'J'
	9-19	SURPFX (N,L,1)	X-direction surface pressure intensity on node 'I'
	20-30	SURPFX (N,L,2)	X-direction surface pressure intensity on node 'J'
	31-41	SURPFY (N,L,1)	Y-direction surface pressure intensity on node 'I'
	42-52	SURPFY (N,L,2)	Y-direction surface pressure intensity on node 'J'

NOTES/

- XXIV (A) (1) Time points for the loading data; Input TD(1) first which represents the starting time point.
- (B) (1) Input the associated nodal points, I, on which the surface pressure is prescribed.
- (2) Input the associated nodal points, J, on which the surface pressure is prescribed.
- (3),(4) Input the surface pressure intensities for bulk load in X and Y directions.
- (5),(6)

XXIV CONTROL CARD FOR SURFACE LOADING DATA INPUT - CONT'D

NOTES/ - CONT'D

XXIV (C) (3),(4), (5),(6) Input the surface pressure intensities for fluid loading in X and Y directions. If NDRN.EQ.0 (undrained/drained case), skip this section entirely.

* All surface loading data cards must be input for as many as 'NSLC' times where 'NSLC' represents the number of surface loading cards for a particular load case.

XXV CONTROL CARDS FOR CONCENTRATED LOADING DATA

* THESE CARDS ARE REQUIRED IN THE EVENT ISLC = 0 (SEE SECTION XXII ON PAGE 355)

A) HEADING CARD IDENTIFYING THE LOADING FUNCTION - (10A8)

Notes	Columns	Variable	Description of Input
(1)	1-80	HED(8)	Title of the loading function

B) INPUT DATA DEFINING LOADED POINTS (I5)

Notes	Columns	Variable	Description of Input
(1)	1-5	NUMLP	Total number of loaded points

NOTES/

XV A (1) Heading card indicating the title of the loading function.

XV B (1) Total number of nodal points where the concentrated loads are prescribed.

XXVI CONTROL CARDS FOR THE HISTORIES OF LOADING
FUNCTION (CONCENTRATED)

- * THESE FOLLOWING SETS OF CARDS i.e. CARD GROUPS (A) & (B) SHOULD COVER THE TOTAL DURATION OF TIME (NCYCL*DT) AS DEFINED IN SECTION IV ON PAGE 336;-

XXVI A) TIME POINTS (8F10.0)

Notes	Columns	Variable	Description of Input
(1)	1-10	TD(N)	Time points for the loading data ie. TD(1), TC(2)

XXVI B) LOADING DATA (I4, 4E10.3)

Notes	Columns	Variable	Description of Input
(1)	1-5	M	Nodal point where the load is prescribed
(2)	6-15	PDYL(N,1,M)	Bulk loading in X - direction
(3)	16-25	PDYL(N,2,M)	Bulk loading in Y - direction
(4)	26-35	PDYL(N,3,M)	Fluid loading in X - direction
(5)	36-45	PDYL(N,4,M)	Fluid loading in Y - direction

NOTES/

- XXVI (A) (1) Time points for the loading data; Input TD(1) first which represents the starting time point.

XXVI CONTROL CARDS FOR THE HISTORIES OF LOADING
FUNCTION (CONCENTRATED) - CONT'D

NOTES/ - CONT'D

- XXVI (B) (1) Input the associated nodal point, sequentially for which the loads are to be prescribed.
- (B) (2) Input the bulk solid and fluid loads
- (3) in the X and Y directions for a particular node; Next repeat the
- (4) loading data for as many nodes up to
- (5) 'NUMLP' as described in section XXV (B) on page 358. Once the input for loading data is finished, repeat XXVI (A) for time sequences ie. TD(2) and continue the data input until the total time duration (NCYCL*DT) is covered.



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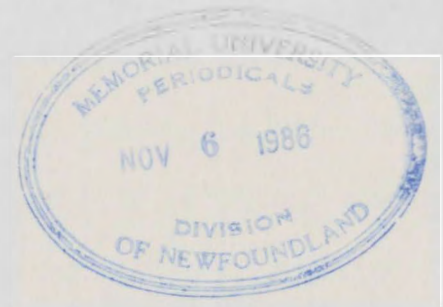
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VOLUME II

A. K. HALDAR



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FINITE ELEMENT DYNAMIC SHAKEDOWN
ANALYSIS OF A GRAVITY TYPE OFFSHORE
STRUCTURE-FOUNDATION SYSTEM

by

(C) A. K. Haldar, B.E., M.Eng, P.Eng.

A thesis submitted to the School
of Graduate Studies in partial
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APPENDIX - E
LISTING OF COMPUTER PROGRAM 'OPFA'

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DATA SET ACORF AT LEVEL 039 AS OF 83/01/17
 SUBROUTINE ACORF(S,C,CT,NEL,NPJ,NPK,AN,NPT)
 IMPLICIT REAL*8 (A-H,O-Z)

DIMENSION AN(NPJ,1),S(NPJ,1),C(NPJ,1)
 DIMENSION SR(8,8),E(8,8),CC(8,8),GM(8)
 DIMENSION P(4,8),D(4,4),LM(8),GH(8),SF(8,8)
 DIMENSION H(8,8),NP(4)

DIMENSION X(16),XN(16),PK(4,8),PN(4,8)

DIMENSION CT(NPJ,1)

DIMENSION ST(20,12),T1(5,12),BK(16),PS(16,8),KM(5),ZM(5)

COMMON/BEINDY/SG(5,200,4,4),NCOND

COMMON/COND/ID,DED(8),KADN,NUMNP,NUMSL,NEG,MBAND,NCYCL,NDYN,IPLNA
 *X,ISTKPK,ISTKST,ISTART,DEGH,MBANDN,NEG5,MBAND5
 COMMON/BOBBY/BKS,NUMGT,FC,LOOP,NDRU,UNIT,LDWA,NGT
 COMMON/ASIR/NYP,NSC,NSLC,NLC,NNYP,UNSC

DIMENSION FACN(4)

*****INITIALIZATION

NDI=1
 IF(NCOND.EQ.1) NDI=4

999 FORMAT(///)
 I0=NEL*3
 M0=10

DO 700 I=1,NPJ
 DO 700 J=1,NPJ
 700 S(I,J)=0.0

DO 800 I=1,NPJ
 DO 820 J=1,NEUS
 C(I,J)=0.0
 CT(I,J)=0.0
 820 CONTINUE

800 CONTINUE

IF(KKS.EQ.4) GO TO 990

DO 970 M0=1,NPJ

00001*33
 00002
 00003**4
 00004**4
 00005**4
 00006*26
 00007**4
 00008*24
 00009*13
 00010**9
 00011*32
 00012*32
 00013**9
 00014*33
 00015*33
 00016*33
 00017*33
 00018*34
 00019*33
 00020**2
 00021**2
 00022*34
 00023*37
 00024*37
 00025*35
 00026*35
 00027*35
 00028*35
 00029
 00030**4
 00031**4
 00032**4
 00033
 00034
 00035
 00036*24
 00037*24
 00038*24
 00039*24
 00040*24
 00041*24
 00042*24
 00043*24
 00044*24
 00045*34
 00046*34
 00047*34
 00048*34
 00049*24
 00050*24
 00051*24
 00052*19
 00053*26
 00054*26
 00055*27

```

DO 960 I=1,NP
960 AUC(I,J)=0.0
970 CONTINUE

```

```

00056*19
00057*26
00058*26
00059*26
00060*26
00061*19

```

```

990 CONTINUE
*****

```

```

JK=3
REWIND 50
REWIND 51
REWIND 12

```

```

00062
00063
00064
00065
00066
00067*18

```

```

REWIND 17

```

```

00068*39
00069*39
00070*18

```

```

REWIND 11

```

```

00071*34
00072*18
00073*18

```

```

***** TAKE 12 PASSED FROM INDEX *****

```

```

00074*18
00075*18
00076*18
00077*18
00078
00079

```

```

I=0
DO 100 J=1,NP

```

```

IF (CODE 1.EQ.0) READ(17) ((D(I,J),J=1,4),I=1,4)
IF (CODE 3.EQ.0) READ(50) ((D(I,J),J=1,4),I=1,4)

```

```

00080*39
00081*39
00082*27

```

```

READ(51) ((D(I,J),J=1,8), (E(I,J),J=1,8), (S(I,J),J=1,8), I=1,8)
*, ((F(I,J),J=1,8), I=1,4), (G(I,J),J=1,8), VOL, DENS, DENS, F, NP

```

```

00083*12
00084*11
00085*21
00086*21

```

```

READ(51) ((S(I,J),J=1,8), I=1,8), ((E(I,J),J=1,8), I=1,8), ((C(I,J),
*,J=1,8), I=1,8), (W(I,J),J=1,8), (H(I,J),J=1,8), I=1,8)
*, (X(I),I=1,16), (Y(I),I=1,16)

```

```

00087*16
00088*14
00089*32
00090*34

```

```

READ(11) ST, TT, PD, BK, ADR, EM, PM, ZM, DS, RT, IFAT, CLR, LMO, YOUNG, PUIS,
*, VOL, ES, FACI

```

```

00091*34
00092*34
00093*34
00094*34

```

```

1200 CONTINUE

```

```

00095
00096

```

```

DO 130 I=1,NP
IF (ACORD.EQ.1) GO TO 1300

```

```

00097*34
00098*34
00099*34

```

```

CALL CORRUTES,P,FACI,IF
WRITE(6,1305)
WRITE(6,1306) ((D(I,J),J=1,8),I=1,4)
WRITE(6,1306) FACI(1F)

```

```

00101*34

```

```

WRITE(6,1309) ((D(I,J),J=1,4),I=1,4)

```

```

1309 FORDAT(SX, H10.3)

```

```

00100*34

```

```

1300 CONTINUE

```

PRINT OF STRAIN DISPL MATRIX

1305	WRITE(6,1305) FORMAT(1H1/5X, ' STRAIN DISPL MATRIX ' /) WRITE(6,1306) ((PK(I,J),J=1,8),I=1,4)	
1306	FORMAT(5X,8F10.3)	
	WRITE(6,1309) ((D(I,J),J=1,4),I=1,4)	
	WRITE(6,1306) VOL	
	DO 200 I=1,4	00102
	*****CODING TO DELETE FOR THIRD ROWS COMPLETELY	00103
	I=1	00104
	IF(I.GT.2) I=I+1	00105
	IF(I1.GT.4) GO TO 200	00106
		00107
		00108
		00109
	I=I+1	00110
	DO 300 J=1,8	00111
		00112*22
	K=INT(J)	00113
	IF(K) 90,90,91	00114
		00115
		00116
91	CONTINUE	00117*26
		00118*33
	IF(NCORD.NE.1) GO TO 1400	
	C(L,K)=P(I1,J)	00119*34
		00120*34
		00121*34
		00122*33
	C(L,K)=P(I1,J)+FACN(IP)	00123*34
	GO TO 90	
1400	C(L,K)=PK(I1,J)	00124*33
	C1(L,K)=PK(I1,J)*VOL	
90	CONTINUE	00125
300	CONTINUE	00126
		00127
		00128
	*****STRESS - STRAIN LAW ASSIGNING FOR ALL ELEMENTS *****	00129
	DO 400 JH=1,4	00130
		00131
		00132*22
	JM=JH	00133
	IF(JH.EQ.3) GO TO 400	00134
	IF(JH.EQ.4) JM=JH-1	00135
		00136*34
	IF(NCORD.NE.1) GO TO 1500	
		00137*34
	JJ=JH+(IP-1)+NSET*(N-1)+NDC+RDI	00138*34

GO TO 1600	00139*34
1500 JJ=JM+(N-1)*NDC+NDI	00140*34
1314 FORMAT(5X,7I5)	
1600 S(I,J)=D(I,I,JI) WRITE(6,1314) JJ,JM,IE,NDC,NCORD,NDI,N	00141*25
400 CONTINUE	00142*34
200 CONTINUE	00143
	00144
	00145
	00146*34
150 CONTINUE	00147*34
*****ELEMENT I ILOC *****	00148
100 CONTINUE	00149
	00150
	00151
1307 WRITE(6,1307) FORMAT(1H1/5X,' GLOBAL STRAIN DISPL MATRIX '/)	
	00152*18
WRITE(6,1310) ((C(I,J),J=1,NEQS),I=1,NPJ)	
WRITE(6,1307)	
WRITE(6,1310) ((CI(I,J),J=1,NEQS),I=1,NPJ)	
WRITE(6,1307)	
WRITE(6,1311) ((S(I,J),J=1,NPJ),I=1,NPJ)	
WRITE(6,1307)	
WRITE(6,1312) ((AN(I,J),J=1,NPK),I=1,NPJ)	
1310 FORMAT(5X,4F10.3)	00153*18
1311 FORMAT(5X,12E10.3)	00154*24
1312 FORMAT(5X,24F5.2)	00155*24
	00156*18
	00157*30
RETURN	00158
END	00159

```
C      PROGRAM A*AIU  
      LPRINT(1) GPAL*8(A=11,11=2J  
      READ(5,100) KKS  
      FORMAT(5X,15)
```

CC 100

```
      IF(KKS.NE.2J) GO TO 200  
      MCO=200000  
      GO TO 300  
      NCOM=400000  
      CONTINUE
```

200
300

C

```
      CONTINUE  
      STOP  
      END
```

```

DATA SET ASSEMB AT LEVEL 009 AS OF 83/02/02
SUBROUTINE ASSEMB(KE,NUMEL,NEO)
IMPLICIT REAL *8 (A-H,O-Z)
DIMENSION RE(1)
DIMENSION LM(16),XP(16)

DO 100 I=1,NEO
  RE(I)=0.0

  **** REWIND TAPE AND READ INFORMATION ****
  REWIND 21
  DO 900 N=1,NUMEL
    READ(21) LM,XP

    WRITE(6,160)
    FORMAT(SX,' *** PRINT IN ASSEMB ***',/)
    WRITE(6,161) (XP(I),I=1,16)
    FORMAT(SX,8E10.3)
    WRITE(6,162) (LM(J),J=1,16)
    FORMAT(SX,16I5)

    DO 360 I=1,16
      L=LM(I)
    400 IF (L) 360,360,200
    200 RE(L)=RE(L)+XP(I)
    360 CONTINUE
    900 CONTINUE
  RETURN
END

```

```

00001**6
00002**2
00003**2
00004**2
00005**2
00006**4
00007**4
00008**6
00009**6
00010**8
00011**3
00012**2
00013**2
00014**2
00015**7
00016**2
00017**2
00018**5
00019**5
00020**8
00021**8
00022**2
00023**2
00024**2
00025**2
00026**2
00027**2
00028**2
00029
00030

```


DATA SET BANDSET	AT LEVEL 002 AS OF 81/09/05	00001
SUERKOHJNF BANDSET (A,B,V,MAXA,NN,NWA,RA,NSCH,DET,ISCALE,KK)		00002
IMPLICIT REAL*(A-H,O-Z)		00003
CALLED BY: SECNTD		00004
COMMON /TAPES/NSTIF,NRED,NL,NR,NT,NMASS		00005
DIMENSION A(NWA),B(I),V(I),MAXA(I)		00006
NR=NN-1		00007
IF (KK-2) 100,700,600		00008
100 TOL=1.0E+04		00009
KTOL=1.0E-07		00010
*** SCALE=2.000**200		00011
SCALE = 1.7000+3H		00012
NTP=3		00013
IS=1		00014
120 REWIND NSTIF		00015
READ (NSTIF) A		00016
DO 140 J=1,NN		00017
140 A(J)=A(J)-KA*B(J)		00018
160 IF (NWA.EQ>NN) GO TO 230		00019
DO 200 M=1,NR		00020
IH=N+NWA-NN		00021
210 IF (A(IH)) 220,215,220		00022
215 IH=IH-NN		00023
GO TO 210		00024
220 MAXA(I)=IH		00025
PIV=A(I)		00026
IF (PIV) 221,500,221		00027
500 IS = IS+1		00028
IF (IS.LE.NTP) GO TO 502		00029
501 WRITE (6,1000) NTP,KA		00030
STOP		00031
502 KA = KA*(1.0-KTOL)		00032
GO TO 120		00033
221 IL=N+NN		00034
L=N		00035
DO 240 J=I,IL,NN		00036
L=L+1		00037
C=A(I)		00038
IF (C) 225,240,225		00039
225 C=C/PIV		00040
IF (DABS(C).LT.TOL) GO TO 235		00041
226 IS=IS+1		00042
IF (IS.LE.NTP) GO TO 245		00043
GO TO 501		00044
245 KA=KA*(1.0-KTOL)		00045
GO TO 120		00046
235 J=L-1		00047
DO 260 K=1,IH,NN		00048
260 A(K+J)=A(K+J)-C*A(K)		00049
A(I)=C		00050
240 CONTINUE		00051
200 CONTINUE		00052
230 IF (A(NN).NE.0.0) GO TO 280		00053
AA=DABS(A(I))		00054
DO 290 I=2,NR		00055
290 AA=AA+DABS(A(I))		00056
		00057
		00058

```

A(NHJ)=-(AA/(K)+1.0E-16)
280 NSCH=0
    ISC=0
    DEL=1.0
    DO 300 I=1,NI
    IF (DABS(DEL).GT. SCALE) GO TO 320
    DEL=DEL/SCALE
    IF (ISC=1)
    320   DEL=DEL+A(I)
    300   IF (A(I).E.O.) NSCH=NSCH+1
    IF (ISCALF.EI.1000) GO TO 340
    ISCALF=ISC
    GO TO 300
340   IF (ISC=SCALE) 350,500,370
    DEL=DEL/SCALE
    GO TO 300
    DEL=DEL+SCALE
    GO TO 340
700   IL=NI
    DO 400 I=1,NI
    CV(I)
    V(N)=C/A(I)
    IF (N/A-N) 410,400,410
    410   IH=MAX(A)
    N=N
    DO 420 I=1,NI,NI
    K=K+1
    V(N)=V(N)-C*(I)
    CONTINUE
    V(NHJ)=V(NHJ)/A(NHJ)
    IF (C/N-A-N) 430,500,430
    430   N=N
    DO 440 I=2,NI
    IH=I*NI
    IH=MAX(A)
    N=N
    DO 460 I=1,NI,NI
    K=K+1
    V(N)=V(N)-A(I)*V(K)
    460   CONTINUE
    440   CONTINUE
    900   CONTINUE
    RETURN
1000  FUNN=I (3700+ERROR) SUBROUTINE STOP IN *BANDLET* / 12X,
1     J(I,13,37H) TRUNCATION FACTORIZATIONS ATTEMPTED, / 12X,
2     IONCORRECT SHIF I = ,E20.14 / 1X)
    END

```

```

00059
00060
00061
00062
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00071
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00073
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00075
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00101
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00108
00109
00110
00111
00112

```

DATA SET CHECK	AT LEVEL 003 AS OF 82/06/14	
SUBROUTINE CHECK(CI,SA,SO,NPJ,NEOS)		00001
IMPLICIT REAL *8 (A-H,O-Z)		00002
DIMENSION C(CI,NPJ),SU(1),SN(1)		00003
DO 300 I=1,NEOS		00004
300 SU(1)=0.0		00005
		00006
		00007
DO 100 I=1,NEOS		00008
DO 200 J=1,NPJ		00009
200 SU(1)=SU(1)+C(I,J,1)*SN(J)		00010
100 CONTINUE		00011
WRITE(6,1000)		00012**2
1000 FORMAT(1H1/' ***** PRINT OF RESIDUAL FORCE IN SYSTEM')		00013**2
WRITE(6,400) (SU(1),I=1,NEOS)		00014**3
400 FORMAT(5X,6E12.5)		00015**3
RETURN		00016
END		00017

DATA SET CUMPGI AT LEVEL, 001 AS OF 82/06/12
SUBROUTINE CUMPGI(PS,F,FACH,IF)

IMPLICIT REAL*8(A-H,O-Z)

DIMENSION PS(10,8),P(4,8),FACH(4)

DO 100 I=1,4

DU 100 I=1,8

JJ=J+(I-1)*4

P(J,I)=PS(IJ,I)

100 CONTINUE

RETURN

END

00001
00002
00003
00004
00005
00006
00007
00008
00009
00010
00011
00012
00013

DATA SET CONST	AT LEVEL 007 AS OF 83/03/02	
SUBROUTINE CONST		00001
IMPLICIT REAL*8 (A-H,O-Z)		00002
COMMON/CONST/DELTA,DELTA2,C1,C2,A0,A01,A02,A03,A04,A05,A06,A07,A08		00003
*X, ISTRER, NPRINT, ISTART, NEQK, NBANDH		00004
* , NEQS, *NBANDS		00005**3
COMMON/ITERGE/ITERA, BETA, GAMA, RTOL, NGAMA, NPUNCH, NEQUIB, ITEMX		00006**3
999 FORMAT(//)		00007
DT=ITERA*DD		00008
		00009
		00010
		00011
*****FINI*****		00012
****ADDED BY HALDAR ****		00013
		00014
100 CONTINUE		00015
		00016
		00017
DELTA=ITERA*DD		00018
DELTA2=DELTA**2		00019
C1=1./TETA		00020
C2=1.-C1		00021
A0=6./DELTA2		00022
A01=3./DELTA		00023
A02=2.*A01		00024
A03=DELTA/2.		00025
A04=A0/TETA		00026
A05=-A02/TETA		00027
A06=1.-3./TETA		00028
A07=DD/2.		00029
A08=DD**2./6.		00030
		00031**7
2100 FORMAT(5X,10E12.5)		00032
RETURN		00033
END		00034

```

DATA SET CONSV      AT LEVEL 036 AS OF 83/01/15
SUBROUTINE CONSV(MAT,EALFA,ELMDA,EMU,IPLNAX,D,ELK,YOUNG,POIS,NEL,
*NG,EM,UMD)
IMPLICIT REAL * 8 (A-H,O-Z)

*****SIS SUBROUTINE CREATES ELEMENT CONSTITUTIVE MATREX BASED
****PROBLEM TYPE . PLANE STRAIN,PLANE STRESS AND AXISYMMETRIC ETC

DIMENSION D(4,4),UMD(4,4)
DIMENSION EALFA(1),ELMDA(1),EMU(1)
***** ADD ADDITIONAL DIMENSION FOR STRAIN COMPTELE AND

COMMON/ASHK/C1,C2,C3,AL,ALF,FACC,FACT
COMMON/BOBBY/KK5,NUMMAT,NC,LOOP,NDRN,NINT,LDWA,NPT
COMMON/ WANCY/ CM(12000)
DIMENSION EM(1)

***** INITIALIZE   *** D MATRIX   *** *****
DO 5200 I=1,4
DO 5200 J=1,4
D(I,J)=0.0
5200 CONTINUE

***** INITIALIZE UNDRAINED CONSTITUTIVE MATRIX *****
DO 360 I=1,4
DO 360 J=1,4
360 UDD(I,J)=0.0

500 CONTINUE
K=MAT
AL=EALFA(K)
ALF=AL+EALFA(K)
FACC=EALFA(K)
FACI=ELMDA(K)

***** PLANE STRESS *****

600 POIS=(ELMDA(K))/(2.+(ELMDA(K)+EMU(K)))
YOUNG=EMU(K)*(2.*(1.+POIS))
CONST=YOUNG/(1.-POIS**2)

IF (IPLNAX) 800,700,700

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```

00001*22
00002*14
00003*20
00004*16
00005*16
00006*12
00007*12
00008
00009
00010
00011*13
00012*13
00013*20
00014*16
00015**2
00016**2
00017**2
00018
00019*32
00020*14
00021*26
00022*14
00023*13
00024*13
00025*13
00026*13
00027*13
00028
00029
00030
00031
00032*14
00033*14
00034*14
00035*14
00036*14
00037*14
00038**3
00039**3
00040**3
00041**2
00042
00043
00044
00045
00046
00047**3
00048**3
00049**3
00050*14
00051*14
00052*14
00053**4
00054**4
00055**4
00056**6
00057*29
00058**6
00059**6

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```

800 CONTINUE
***** CHECK WHAT HAPPEN IF WE PUT C1, C2, ... DIRECTLY
***** D(I,J) FOR FLARE STRESS CONDITION AND
***** FLARE STRAIN OR AXISYMMETRIC *****
700 CONTINUE
530 CONTINUE
C1=Z.0*E*U(K)+P*AC1
C2=FACT
C3=E*U(K)
C4=E*(P*U(K)+(Z.0*E*U(K))/J.)
BLK=C4
900 CONTINUE
***** ELASTIC ISOTROPIC CONSTITUTIVE MATRIX *****
D(1,1)=C1
D(1,2)=C2
D(1,3)=D(1,2)
D(3,3)=D(1,1)
D(2,2)=C1
D(2,3)=D(1,3)
D(4,4)=C3
540 CONTINUE
DO 520 I=1,4
DO 520 J=1,4
D(I,J)=D(I,J)
520 CONTINUE
IF (JPLD).OR.(-1) GO TO 400
***** COMPENSATE FOR ZERO STRESS CONDITION IN Z DIRECTION
D(1,1)=D(1,1)-D(3,1)*D(1,3)/D(3,3)
D(1,2)=D(1,2)-D(3,2)*D(1,3)/D(3,3)
D(1,4)=D(1,4)-D(3,4)*D(1,3)/D(3,3)
D(2,2)=D(2,2)-D(3,2)*D(2,3)/D(3,3)
D(2,4)=D(2,4)-D(3,4)*D(2,3)/D(3,3)
D(4,4)=D(4,4)-D(3,4)*D(4,3)/D(3,3)
DO 650 I=1,4
DO 680 J=1,4
680 D(I,J)=D(I,J)
D(1,3)=0.0

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```

00060**6
00061**6
00062*12
100063*12
00064*12
00065*12
00066*12
00067**3
00068**3
00069**3
00070**2
00071*26
00072*26
00073*27
00074*26
00075*26
00076
00077
00078
00079**7
00080**7
00081**6
00082**6
00083**2
00084**3
00085**3
00086**3
00087**3
00088
00089
00090
00091
00092
00093
00094
00095*27
00096*27
00097*27
00098
00099
00100
00101
00102*12
00103*12
00104*12
00105*12
00106*12
00107*12
00108*12
00109*12
00110*12
00111*12
00112*12
00113*12
00114*12
00115*12
00116*12
00117*13
00118*13
00119*12

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```

650 D(3,1)=0.0
400 CONTINUE
***** MULTIPLY CONSTITUTIVE MATRICES *****
DN=ALF*E*(N)
UDD(1,1)=D(1,1)+DN
UDD(2,2)=D(2,2)+DN
UDD(1,2)=D(1,2)+DN
UDD(2,1)=UDD(1,2)
UDD(1,3)=UDD(1,2)
UDD(3,3)=UDD(1,1)
UDD(2,3)=UDD(1,3)
UDD(4,4)=D(4,4)
UDD(3,1)=UDD(1,3)
UDD(3,2)=UDD(2,3)
***** REMOVE UNDRAINED POIS AND YOUNG *****
WRITE(50) ((UDD(I,J),J=1,4),I=1,4)
300 WRITE(17) ((D(I,J),J=1,4),I=1,4)
***** CONNECT HERE FOR ELASTO PLASTIC ANALYSIS FROM ELPAL
350 CONTINUE
250 CONTINUE
RETURN
END
00120*33
00121*33
00122*12
00123*12
00124*12
00125*13
00126*13
00127*13
00128*13
00129*14
00130*14
00131*14
00132*14
00133*14
00134*14
00135*14
00136*14
00137*14
00138*23
00139*23
00140*14
00141*14
00142*14
00143*25
00144*25
00145*14
00146*14
00147*34
00148*14
00149*13
00150*13
00151*36
00152*35
00153*34
00154*12
00155*12
00156*12
00157*12
00158*12
00159*14
00160*14
00161*17
00162*17
00163*17
00164*28
00165*28
00166*28
00167
00168

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```

      DATA SET CURV52      AT LEVEL 001 AS OF 81/07/23
SUBROUTINE CURV52(S,GN,DI)      00001
IMPLICIT REAL*8(A-H,O-Z)      00002
DIMENSION GCLAY(11),DCLAY(11) 00003
DATA GCLAY/1.00,0.913,0.761,0.565,0.400,0.261,0.152,0.076,0.037, 00004
1 0.013,0.004/                00005
DATA DCLAY/2.50,2.50,2.50,3.50,4.75,6.50,9.25,13.8,20.0,26.0,29.0/00006
S=DABS(S)                      00007
S=DLG10(S)+2.+9.                00008
K=S                             00009
IF(K.GE.1) GO TO 2              00010
GN=1.                           00011
DN=DCLAY(1)                     00012
RETURN                          00013
2 IF(K.LY.11) GO TO 3           00014
GN=GCLAY(11)                    00015
DN=DCLAY(11)                    00016
RETURN                          00017
3 GN=GCLAY(K)+(GCLAY(K+1)-GCLAY(K))*(S-K) 00018
DN=DCLAY(K)+(DCLAY(K+1)-DCLAY(K))*(S-K) 00019
RETURN                          00020
END                              00021

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DATA SET DEQJIK      AT LEVEL 037 AS OF 63/03/02
SUBROUTINE DEQJIK(X0,NDYN,A0,RE,R,LTR,NITR,ITEMX,RTOL,NC,A4,NEQ, 00001
                                00002
*MBAND, NWA, NDOB, NDBLOCK, XI, MAXA, XZ, X1, VEL, AC, B, ID, AMAX, PMAX, AM, EK, AN 00003**7
                                00004
*, ANI, EKJ, EM, NUMNP, NUMEL, RNORF, XM, A5, GB, AT, BT, 00005*16
                                00006*16
*AS, LCJ 00007*16
                                00008
IMPLICIT REAL *B(A-H,U-Z) 00009
                                00010
                                00011
COMMON/GORA/IFLAG, IGR 00012*22
COMMON/ASIM/NYP, NSC, NSIC, NLC, NNYP, NNSC, ISLC 00013
                                00014*14
COMMON/ACNST/DELTA1, DELTA2, C1, C2, A0, A01, A02, A03, A04, A05, A06, A07, A 00015*14
*08 00016*14
COMMON/CUTL/DD, DEL(B), RADN, NUMNG, NUMSL, NGU, MBDN, NCYCL, NDN, 00017*30
*IPLNAX, ISIRPE, NPRINT, ISIAST, NEQH, MBANDH 00018*30
*, NEQS, MBANDS 00019*31
                                00020*31
                                00021*30
COMMON/BEJA/NS, NELYF, ND, NUMGT, MODEL, NCON, IDW, NK015, NK016, NK017

                                00022*14
                                00023*14
DIMENSION X0(1), A0(1), RE(1), R(1), A4(NEQ,1), MAXA(1) 00024
DIMENSION A3(NEQ,1) 00025*16
                                00026
DIMENSION AT(1), BT(1), GB(1) 00027*11
COMMON/BOBBY/RKB, NOMBAT, KC, DDBP, NDRN, NINT, IDWA, NPT 00028*29
                                00029**7
                                00030
DIMENSION X1(1), X2(1), VEL(1), AC(1), H(1), ID(NUMNP,1), AMAX(NUMNP,2), 00031
*2), PMAX(NUMEL,1), AM(1), EK(1), AN(NNSC,1), ANI(NYP,1), EKJ(1), DS(4,3) 00032**4
DIMENSION XM(1), A5(NEQ,1) 00033**5
                                00034*12
                                00035*12
COMMON/HARVEY/UACC(700), UVEL(700), UDISPL(700) 00036*35
COMMON/TAPES/NUM(6), NUMBER, LCOUNT

                                00037*12
                                00038*35
DO 150 I=1,NEQ 00039*34
UACC(I)=0.0 00040*34
UVEL(I)=0.0 00041*34
UDISPL(I)=0.0 00042*34
150 CONTINUE 00043*12
                                00044*14
***** CONVERSION OF INTEGRATION OF CONSTANTS. ***** 00045*14
***** 00046*14
A0 =A0 00047*14
A01=A01 00048*14
A02=A2 00049*14
A03=A5 00050*14
A04=A6 00051*14
A05=A7 00052*14
A06=A8 00053*14
A07=A9 00054*14
A08=A10

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*****
00055*14
00056*14
00057*14
00058
00059**3
00064**4
00065**4
00066**4
00067**4

IF(NC.EQ.0) GO TO 31
00068**4
IF(LCOUNT.NE.JSTRPR) GO TO 32
00069**4
00070**4

31 CONTINUE

152 WRITE(6,152)
    FORMAT(1H1/5X,' PRINT IN DEQIR MASS,STIFFNESS,DAMPING ')
153 WRITE(6,153) (XM(I),I=1,NEQ)
    FORMAT(5X,1ZE10.3)
154 WRITE(6,154) ((A5(I,J),J=1,MBAND),I=1,NEQ)
    FORMAT(5X,8E10.3)

WRITE(6,65) NC,ITR
WRITE(6,66) (AU(I),I=1,NEQ)
65 FORMAT(1H1/5A,' PRINT IN DEQIR ' /
   *5X,' NO. OF TIME STEP = 1,15/
   *5X,' NO. OF ITERATION = 1,15/
   *5X,' DISPL. ARRAY AU BEGINING ITERATION /)

66 FORMAT(5X,8E10.3)

32 CONTINUE
00071*16
00072*16
00073*16
00074*16
00075*16
00076*16
00077*16
00078*16
00079*16
00080*15
00081**4

DO 50 ITR=2,NITR
00082**4
00083**4
00084**4
00085*14

DO 60 I=1,NEQ
00086*14
00087*14
00088*14
00089*12
00090*35
00091*35
00092*35
00093*12
00094*12
00095*35
00096*35
00097*35

IF(NDYN.EQ.4) GO TO 55
00098*12
00099*12
00100*12
55 CONTINUE
    IF(ITR.NE.2) GO TO 60
    UDISPL(1)=X0(1)+AU(1)
60 CONTINUE

    IF(NC.EQ.0) GO TO 52
    00101*12
    IF(LCOUNT.NE.JSTRPR) GO TO 53
    00102**4

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52 CONTINUE
WRITE(6,*)
WRITE(6,66) (DISPL(I), I=1, NEQ)
WRITE(6,66) (VEL (I), I=1, NEQ)
WRITE(6,66) (UACC (I), I=1, NEQ)
WRITE(6,66) (X0 (I), I=1, NEQ)
WRITE(6,66) (X1 (I), I=1, NEQ)
WRITE(6,66) (X2 (I), I=1, NEQ)
67 FORMAT(5X, ' (I-1) TH. STEP ITERATION VALUES FOR DISPL. ')
WRITE(6,101) ND, WELTYP, ND, NUMGI, MODEL, NCON, IDW, NK015, NK016, NK017
101 FORMAT(5X, 10I5)
53 CONTINUE
IGR=ITR
CALL DMPL(X0, X2, VEL, AC, AD, B, ID, DS, AMAX, FMAX, NUMNP, NUME
*L, NC, AM, EN, AN, AN1, ENJ, TM, LC, NEG, RE, ITR)
WRITE(6,66) (UACC(I), I=1, NEQ)
CALL ASSEMBL(PE, NUMEL, NEG)
REWIND 21
IF(NC, EQ, 0) GO TO 51
IF(LCOUNT.NE.1STRPK) GO TO 56
51 CONTINUE
WRITE(6,68)
68 FORMAT(10I5, ' RESIDUAL LOAD VECTOR AT (I-1) STEP ')
WRITE(6,66) (RE(I), I=1, NEQ)
56 CONTINUE
***** CALCULATE OUT OF BALANCE LOADS FOR THIS ITERATION ****
DO 390 I=1, NEQ
390 R(I)=GR(I)-RE(I)
DO 391 I=1, NEQ
391 RE(I)=0.0
WRITE(6,66) (R(I), I=1, NEQ)
GO TO (1, 1, 3, 3, 3), MDYN
1 CONTINUE
***** CALCULATE THE EFFECT OF ACCLN, VEL, DISPL AT PREVIOUS ST00143*11
00103*16
00104*16
00105*16
00106*16
00107*16
00108*16
00109*16
00110*16
00111*16
00112*14
00061**4
00062*22
00063*22
00113*14
00114*18
00115*18
00116*18
00117*14
00118*14
00119**8
00120**4
00121*16
00122*16
00123*16
00124*16
00101*12
00102**8
00125*16
00126*16
00127*18
00128*16
00129*16
00130*16
00131**4
00132
00133
00134
00135
00136*14
00137*11
00138
00139*11
00140*11
00141*11
00142*11
00143*11
00144*12

```

	WRITE(6,66) (DACC(1),I=1,NEQ)	
	WRITE(6,66) (AM(1),I=1,NEQ)	
	DO 61 I=1,NEQ	00145*12
	RE(I)= RE(I)+AM(I)*DACC(I)	00146*14
61	CONTINUE	00147*12
	WRITE(6,66) (RE(I),I=1,NEQ)	
		00148*12
155	DO 155 I=1,NEQ	
	R(I)=R(I)-RE(I)	
	WRITE(6,66) (R(I),I=1,NEQ)	
	DO 156 I=1,NEQ	
156	RE(I)=0.0	
	DO 157 I=1,NEQ	
157	RE(I)= RE(I)+A5(1,I)*UVEL(I)	00146*14
		00149*12
	DO 460 I=1,NEQ	00150*12
	LQ=I-1	00151*12
	MR=MINO(MBAND,NEQ-I+1)	00152*12
	IF(MR.EQ.1) GO TO 485	00153*12
	DO 470 J=2,MR	00154*12
	K=LQ+J	00155*12
	RE(I)=RE(I)+A5(1,J)*HVEL(K)	00156*14
	RE(K)=RE(K)+A5(1,J)*HVEL(I)	00157*14
		00158*14
		00159*14
470	CONTINUE	00160*12
480	CONTINUE	00161*12
485	CONTINUE	00162*12
		00163*12
	WRITE(6,66) (RE(I),I=1,NEQ)	
	DO 158 I=1,NEQ	
158	R(I)=R(I)-RE(I)	
	WRITE(6,66) (R(I),I=1,NEQ)	
		00164*12
		00165*11
3	CONTINUE	00166*11
	IF(NC.EQ.0) GO TO 57	
		00101*12
	IF(LCOUNT.NE.1STRPR) GO TO 58	
		00102*10
57	CONTINUE	
	WRITE(6,69)	00167*11
69	FORMAT(5X,' UNBALANCED LOAD VECTOR ANYWAY 1/')	00168*16
	WRITE(6,66) (R(I),I=1,NEQ)	00169*16
		00170*17
		00171*16
58	CONTINUE	
		00172*11
		00173
	IF(1TR.LE.(ITEMX/2+2)) GO TO 201	00174*18
	** CALCULATE NORM OF OUT OF BALANCE LOADS AND CHECK FOR DIVERGENCE	00175
		00176
	RENORM=0.	00177
	DO 410 I=1,NEQ	00178
		00179*24
410	RENORM=RENORM+R(I)*R(I)	00180*24

		00181*24
		00182*26
	IF(REFORM.EQ.RWORK) GO TO 201	00183*26
		00184
		00185*24
		00186*24
		00187*14
	WRITE(6,2050) IPR	00188
	RETURN	00189
		00190
		00191
		00192
201	CONTINUE	00193*10
	IF(NC.EQ.0) GO TO 200	00194**8
		00195**8
	CALL SYMSOL(1,A3,R,NEG,MBAND)	00196*14
	CALL SYMSOL(2,A3,R,NEG,MBAND)	00197*14
		00198*14
		00199*12
		00200*12
	GO TO 400	00201*12
200	CONTINUE	00202*12
		00203**9
		00204**8
		00205**4
	CALL REDVK(A4,R,MAXA,NEQB,NWA,NEG,NBLUCK,MI,MBAND,NC,KKS)	00206*11
		00207*14
		00208*11
460	CONTINUE	00209*11
		00210*11
		00211*11
		00212*11
		00213**4
		00214
		00215
		00216
	***** UPDATE DISPL. FOR CURRENT ITERATION STEP *****	00217*14
	DO 100 I=1,NEG	00218*14
	AD(I)= AD(I)+R(I)	00219*14
	UDISPL(I)= AD(I) +R(I)	00220*14
100	CONTINUE	00221*14
	IF(NC.EQ.0) GO TO 59	
		00101*12
	IF(LCOUNT.NE.IDTRPR) GO TO 19	
		00102**8
59	CONTINUE	
		00222
	WRITE(6,71)	00223*16
71	FORMAT(5X,' INCREMENTAL R(I) DISPL. VECTOR FOR THIS ITER. ')	00224*16
	WRITE(6,66) (R(I),I=1,NEG)	00225*16
	WRITE(6,79)	00223*16
79	FORMAT(5X,' INCREMENTAL AD(I) DISPL. VECTOR FOR THIS ITER. ')	00224*16
	WRITE(6,66) (AD(I),I=1,NEG)	00225*16
	WRITE(6,72)	00226*16
72	FORMAT(5X,' UPDATED DISPL. COMP. AD(I) VECTOR ')	00227*16
	WRITE(6,66) (UDISPL(I),I=1,NEG)	00228*16
		00229*16
19	CONTINUE	
		00230*16
		00231

**** CHECK FOR CONVERGENCE ****

DNORM=0.
DINORM=0.
DO 420 I=1,NEQ
DNORM=DNORM+UDISPL(I)*UDISPL(I)

420 DINORM=DINORM+R(I)*R(I)
TOL=DNORM*RTOL

IF(DINORM.LE.TOL) GO TO 430
WRITE(6,2011) DINORM, DNORM, RTOL, TOL
2011 FORMAT(5X, 'E10.3')

IF(ITR.LT.ITEMX) GO TO 50
WRITE(6,2016) NC, ITR
WRITE(6,2020)
STOP

50 CONTINUE

430 CONTINUE

DO 99 I=1,NEQ
AU(I)=UDISPL(I)-AU(I)
99 CONTINUE

IF(NC.EQ.0) GO TO 81

IF(LCOUNT.NE.1STRPK) GO TO 62

81 CONTINUE

WRITE(6,73)

73 FORMAT(1H1/5X, ' * ***** NEW DISPL. INC. U,I THIS ITR. ** '/')

WRITE(6,74) (AU(I), I=1,NEQ)

74 FORMAT(5X, 'E10.3')

WRITE(6,75) NC, ITR, DNORM, DINORM

75 FORMAT(5X, ' NC = ', 15/
* 5X, ' ITR = ', 15/
* 5X, ' DNORM = ', E10.3/
* 5X, ' DINORM = ', E10.3/)

62 CONTINUE

00232
00233
00234
00235
00236
00238*36
00239*36
00240*36
00241*14
00242*33
00243
00254*34
00255
00256*23

00257*23
00258
00259
00260
00261
00262
00263
00264
00265
00266
00267
00268
00269
00270
00271**4
00272**3
00273**3
00274*12
00275*12
00276*15

00101*12
00102**4

00244*34
00245*34
00246*34
00247*34
00248*34
00249*34
00250*34
00251*34
00252*34
00253*34
00277*14

00278*33

	GO TO (2,2,4,4,4),DDYD	00279*33
4	CONTINUE	00280*33
		00281*33
		00282*33
5	DO 5 I=1,NEQ X0(I)=00DISPL(I)	00283*18
		00284*18
		00285*18
		00286*18
	GO TO 8	00287*14
		00288*33
		00289*33
2	CONTINUE	00290*14
		00291*15
		00292*15
		00293*15
		00294*14
	***** CALCULATE CURRENT CONVERGED ACCLN,VEL DISPL,ETC ***	00295*14
		00296*18
		00297*18
	DEL1=DT	00298*18
	DO 7 I=1,NEQ	00299*18
	U0=X0(I)	00300*18
	U10=X1(I)	00301*18
	U20=X2(I)	00302*18
	U=A0(I)	00303*18
	X2(I)=A04*U+A05*U10+A06*U20	00304*18
	X1(I)=U10+DEL1/Z.*(U20+X2(I))	00305*18
7	X0(I)=U0+DEL1*U10+A08*(X2(I)+Z.*U20)	00306*18
	CONTINUE	00307*18
		00308*18
		00309*18
8	CONTINUE	00310*14
		00311*33
		00312*33
		00313*14
		00314*15
		00315*15
		00316*14
		00317*14
	***** CONVERT HERE CURRENT INC.DISPL(CONVERGED) TO TOTAL AT	00318*12
		00319*12
		00320*12
	WRITE(6,2012)	
2012	FORMAT(1H1/5X,1 CONVERGED VALUES OF DISPL,VEL,ACCN IN DEG/TH /)	
	WRITE(6,2013) (X0(I),I=1,NEQ)	
	WRITE(6,2013) (X1(I),I=1,NEQ)	
	WRITE(6,2013) (X2(I),I=1,NEQ)	
2013	FORMAT(5X,12E10.3)	
	RETURN	00321**2
		00322**2
2010	FORMAT(//// 5TH EQUILIBRIUM ITERATION IN TIME STEP = ,15//	00323**2
*	3TH NUMBER OF ITERATIONS = ,15 /)	00324**2
2020	FORMAT(////46H ITERATION LIMIT REACHED S T O P OF SOLUTION)	00325**2
2050	FORMAT(////70H OUT-OF-BALANCE LOADS LARGER THAN INCREMENTAL LOADS	00326**5
	*AFTER ITERATION = ,15/)	00327**5
		00328**2
END		00329**2


```

DATA SET DRUCK      AT LEVEL 022 AS OF 83/02/16
SUBROUTINE DRUCK(FRUF,SIG,EPS,ANGLE,CUDES,IPEL,PF,PS,FACN,LM, 00001*19
*PMAX,NUMEL)      00002*19
IMPLICIT REAL*8(A-H,I-Z) 00003*18
                                00004
                                00005
                                00006
                                00007
                                00008
                                00009
                                00010
                                00011
                                00012
                                00013
                                00014
                                00015
                                00016
                                00017
                                00018
                                00019
                                00020
                                00021
                                00022
                                00023
                                00024
                                00025
                                00026
                                00027
                                00028**2
                                00029**2
                                00030*18
                                00031
                                00032
                                00033**2
                                00034**2
                                00035*17
                                00036*16
                                00037**2
                                00038**2
                                00039**2
                                00040*17
                                00041*19
                                00042*19
                                00043*17
                                00044*17
                                00045*19
                                00046*17
                                00047**2
                                00048*19
                                00049*19
                                00050**2
                                00051*17
                                00052*17
                                00053
                                00054**2
                                00055
                                00056
                                00057**2
                                00058**2
                                00059**2

IS1      NUMBER OF STRESS COMPONENTS
ISK      NUMBER OF STRAIN COMPONENTS
SIG      STRESSES AT THE END OF THE PREVIOUS UPDATE
EPS      STRAINS AT THE END OF THE PREVIOUS UPDATE
RATIO   PART OF STRAIN INCREMENT TAKEN ELASTICALLY
DELEPS  INCREMENT IN STRAINS
DELSIG  INCREMENT IN STRESSES, ASSUMING ELASTIC BEHAVIOR

PROP(1)  YOUNG S MODULUS
PROP(2)  POISSON S RATIO
PROP(3)  ANGLE OF FRICTION (INPUT UNIT = DEGREES)
PROP(4)  COHESION

IPEL    = 1, MATERIAL ELASTIC,
        = 2, MATERIAL PLASTIC

COMMON/EL/IND,ICOUNT
COMMON/VAR/NG,KPRI
COMMON/MATMOD/STRESS(9), STRAIN(4),C(4,4),IPT,NEL
COMMON /DKPRAG/ ISK,IST,A1,B1,C1,A2,B2,C2,D2,G,BM,ALFA,YLD
COMMON /DKPRAG/ A1,B1,C1,A2,B2,C2,D2,G,BM,ALFA,YLD,ISK,IST
COMMON/BELA/NS,NELTYP,ND,NUNGT,MODEL,NCON,IDW,NK015,NK016,NK017
COMMON/SORIN/IT(5,12),XP(16),INDMB,ITP2D
COMMON/INTEGE/FEIA,BEIA,GAMA,KFDL,NGAMA,NPUNCH,NEQUIB,ITEMX
COMMON/CONTL/ DO,HEB(B),KADN,NUMMP,NUMSL,NEQ,MBAND,NCYCL,NDYN,IPLN
*AX,ISTRPR,NPRINT,ISTART,NEQH,MBANDH

COMMON/FRIM/ISTRES,MPRINT,JPRINT,KPRINT,IP
COMMON/FUBA/ABA,FM,EPSV

COMMON/KALI/OP1,KFL,JPLDT(200),REDMT(200)

COMMON/BOBBI/KKS,NUMMAT,NC,LOOP,NDRN,NINT,IDWA,NPT

COMMON/GOKA/IFLAG,ITR

DIMENSION PMAX(NUMEL,1)

DIMENSION PROP(NCON),SIG(4),EPS(4)
DIMENSION TAU(4),DELSIG(4),DELEPS(4),DEPS(4),STATE(2)

EQUIVALENCE (DELEPS(4),DEPS(4))

DIMENSION IT(1)

```

```

DIMENSION PS(16,8),FACN(4),LM(16)
DIMENSION GSIRES(9)
DATA NGLAST/1000/, STATE/INE,IMP/
1
TIME=IND*DD
00060
00061*14
00062*19
00063
00064
00065
00066*17
00067*17
00068
00069
00070
00071
00072
00073
00074
00075
00076
00077
00078
00079
00080
00081
00082
00083
00084
00085*17
00086*17
00087*17
00088
00089
00090
00091
00092
00093
00094
00095
00096
00097
00098
00099
00100
00101
00102
00103
00104
00105
00106
00107
00108
00109
00110
00111
00112
00113
00114
00115
00116
00117
00118
00119

DIMENSION PS(16,8),FACN(4),LM(16)
DIMENSION GSIRES(9)
DATA NGLAST/1000/, STATE/INE,IMP/
1
TIME=IND*DD

IF (IPT.NE.1) GO TO 110

ISF=4
IF (I1YP2D.EQ.2) ISF=3
ISR=3
IF (I1YP2D.EQ.0) ISR=4
YM=PRDP(1)
PV=PRDP(2)
D2=PV/(PV-1.)
G =YM/(1.+PV)/2.
BM=YM/(1.-2.*PV)/3.
A2=BM+4.*G/3.
B2=BM-2.*G/3.
C1=G
C2=G

***** CHANGE FOR UNDRAINED CHARACTERISTIC *****

IF (I1YP2D.EQ.2) GO TO 105
PLANE STRAIN / AXISYMMETRIC
A1=A2
B1=B2
GO TO 110

PLANE STRESS
105 A1=YM/(1.-PV*PV)
B1=A1+PV

110 ANG=ANGLE
CFE=CHIEF
YM=DCOS(ANG)
PV=DSIN(ANG)
YLD=1.732050808*(3.-PV)

ALFA=2.*PV/YLD
YLD=6.*CFE*YM/YLD

1. CALCULATE INCREMENTAL STRAINS
DO 120 I=1,ISR
120 DELEPS(I) = STRAIN(I) - EPS(I)

2. CALCULATE THE STRESS INCREMENT,
ASSUMING ELASTIC BEHAVIOR
DELSIG(1) = A1*DELEPS(1) + B1*DELEPS(2)
DELSIG(2) = B1*DELEPS(1) + A1*DELEPS(2)

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DELSIG(3) = (1*DELEPS(3)) 00120
DELSIG(4) = 0. 00121
IF (I1YP2D.EQ.2) GO TO 150 00122
DELSIG(4) = B1 * (DELEPS(1)+DELEPS(2)) 00123
IF (I1YP2D.EQ.1) GO TO 150 00124
DELSIG(1) = DELSIG(1) + B1*DELEPS(4) 00125
DELSIG(2) = DELSIG(2) + B1*DELEPS(4) 00126
DELSIG(4) = DELSIG(4) + B1*DELEPS(4) 00127
00128
3. CALCULATE TOTAL STRESSES, 00129
ASSUMING ELASTIC BEHAVIOR 00130
00131
150 TAU(4) = 0. 00132
DO 160 I=1,1SF 00133
160 TAU(I) = SIG(I) + DELSIG(I) 00134
00135
4. CHECK WHETHER *TAU* STATE OF STRESS FALLS 00136
OUTSIDE THE LOADING SURFACE 00137
00138
SM = (TAU(1)+TAU(2)+TAU(4))/3. 00139
SX = TAU(1) - SM 00140
SY = TAU(2) - SM 00141
SZ = TAU(4) - SM 00142
SS = TAU(3) 00143
SZ = TAU(4) - SM 00144
SBAR=DSQRT( .5*(SX*SX+SY*SY+SZ*SZ) +SS*SS) 00145
00146
F1=3.*ALFA+SM + SBAR - YLD 00147
00148
IF (F1) 170,170,300 00149
00150
STATE OF STRESS WITHIN LOADING SURFACE - ELASTIC BEHAVIOR 00151
00152
170 IPEL=1 00153
STRESS(4) = 0. 00154
DO 180 I=1,1SF 00155
180 STRESS(I) = TAU(I) 00156
IF (I1YP2D.EQ.2) STRAIN(4)=EPS(4) + D2*(DELEPS(1)+DELEPS(2)) 00157
GO TO 400 00158
00159
STATE OF STRESS OUTSIDE LOADING SURFACE - PLASTIC BEHAVIOR 00160
00161
300 IF (IPEL.EQ.1) GO TO 320 00162
00163
.....WAS PLASTIC 00164
00165
IPEL=2 00166
RATIO = 0. 00167
DO 315 I=1,1SF 00168
315 TAU(I) = SIG(I) 00169
GO TO 370 00170
00171
.....WAS ELASTIC 00172
00173
DETERMINE PART OF STRAIN TAKEN ELASTICLY 00174
00175
320 IPEL=2 00176
00177
SM = (SIG(1)+SIG(2)+SIG(4))/3. 00178
00179

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SX = SIG(1) - SM          00180
SY = SIG(2) - SM          00181
SS = SIG(3)              00182
SZ = SIG(4) - SM          00183
DM = (DELSIG(1)+DELSIG(2)+DELSIG(4))/3. 00184
DX = DFELSIG(1) - DM      00185
DY = DFELSIG(2) - DM      00186
DS = DFELSIG(3) - DM      00187
DZ = DFELSIG(4) - DM      00188
A=DX*DX + DY*DY + 2.*DS*DS + DZ*DZ - 18.*(ALFA*DM)**2 ) 00189
B=SX*DX + SY*DY + 2.*SS*DS + SZ*DZ + ALFA*DM*(6.*YLD-18.*ALFA*SM) 00190
E=SA*SX + SY*SY + 2.*SS*SS + SZ*SZ + 00191
1 ALFA*SM*(12.*YLD-18.*ALFA*SM) - 2.*YLD*YLD 00192
RATIO=(-B + DSQRT(B*B-A*E))/A 00193
DU 350 I=1,1ST 00194
350 TAU (1) = SIG(1) + RATIO*DELSIG(1) 00195
IF (IYF2D.EQ.2) STRAIN(4)=EPS(4)+RATIO*D2*(DELEPS(1)+DELEPS(2)) 00196
*TAU* NOW CONTAINS (PREVIOUS STRESSES + 00197
STRESSES DUE TO ELASTIC STRAIN INCREMENTS) 00198
5. CALCULATE PLASTIC STRESSES 00199
DETERMINE INCREMENT INTERVAL. 00200
M=20.*DSQRT(2.*YLD*E)/YLD+1. 00201
IF (M.GT.30) M=30 00202
XM = (1.-RATIO)/FLOAT(M) 00203
DU 380 I=1,1SK 00204
380 DEFS(I) = XM*DELEPS(I) 00205
..... CALCULATION OF ELASTOPLASTIC STRESSES ..... (START) 00206
DU 600 IM=1,M 00207
CALL PRAGER (TAU,DEFS,C) 00208
DU 560 I=1,1ST 00209
DU 560 J=1,1SK 00210
560 TAU(J) = TAU(1) + C(1,J) * DEFS(J) 00211
CORRECTION 00212
DM = (TAU(1)+TAU(2)+TAU(4))/3. 00213
DX = TAU(1) - DM 00214
DY = TAU(2) - DM 00215
DS = TAU(3) 00216
DZ = TAU(4) - DM 00217
PERFECTLY PLASTIC MATERIAL 00218
SBAR=DSQRT(.5*(DX*DX+DY*DY+DZ*DZ)+DS*DS) 00219

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F1=3.*ALFA*DM + SBAR - YLD          00240
FIA=DAHS(F1)/YLD                    00241
IF (FTA.IF,0.005) GO TO 600         00242
                                     00243
A=.25 - 9.*(ALFA**4)                00244
B=SBAR - 18.*(ALFA**3)*DM + 6.*(ALFA**2)*YLD 00245
E=(SBAR**2) - 9.*(ALFA**2)*(DM**2) + 6.*ALFA*YLD*DM - YLD*YLD 00246
E=(B-DSQR4(B**2-4.*A*E))/(2.*A)     00247
                                     00248
TAU(1)=IAU(1) - E*(ALFA + .5*DX/SBAR) 00249
TAU(2)=IAU(2) - E*(ALFA + .5*DY/SBAR) 00250
TAU(3)=IAU(3) - E*(.5*DS/SBAR)       00251
IF (IST.FU,3) GO TO 600             00252
TAU(4)=IAU(4) - E*(ALFA + .5*DZ/SBAR) 00253
                                     00254
600 CONTINUE                         00255
                                     00256
..... CALCULATION OF ELASTOPLASTIC STRESSES ..... ( END ) 00257
                                     00258
STRESS(4) = 0.                      00259
DO 390 I=1,1SI                       00260
390 STRESS(I) = TAU(I)                00261
                                     00262*14
400 CONTINUE                          00263*19
                                     00264*19
IF (IFLAG.EQ.0) GO TO 610            00265*19
                                     00266*14
                                     00267
UPDATING STRESSES, STRAINS, ANGLE, COHESION 00268
                                     00269
DO 410 I=1,1SI                       00270
410 SIG(I) = STRESS(I)                00271*19
DO 420 I=1,1SK                       00272
420 EPS(I) = STRAIN(I)                00273
IF (IYFZD.EQ.2) EPS(4)=STRAIN(4)    00274
ANGLE=ANG                             00275
COHES=CEF                             00276
                                     00277
610 CONTINUE                          00278
                                     00279*14
                                     00280
7. FORM THE MATERIAL LAW              00281
IF (IPED.EQ.1) GO TO 450              00282
                                     00283
ELASTO-PLASTIC                        00284
CALL PRAGER (STRESS,DEFD,C)          00285
                                     00286
                                     00287
                                     00288**3
                                     00289**3
                                     00290**3
                                     00291**3
PG=PF(IPT)                             00292*12
CALL PRSTS(STRESS,PG,GSTRES)          00293*12
CALL MODCOM(C,NEH,PG,GSTRES)          00294*12
                                     00295*12
                                     00296*12
IF (IFLAG.NE.0) GO TO 620            00297*14
                                     00298*14
                                     00299*14

```

```

CALL ITRESLD(NEL,IP1,GSIRES,FACN,PS,TT,PC,LM,XP)
RETURN
620 CONTINUE.

      IF (IND) 820,700,820
820 CONTINUE

IF (ISTRES.EQ.1STPR) GO TO 700
RETURN
ELASTIC
450 DO 460 I=1,1ST
DU 460 J=1,1ST
460 C(I,J)=0.
C(1,1)=A1
C(2,1)=B1
C(1,2)=B1
C(2,2)=A1
C(3,3)=C1

***** CHANGE HERE FROM ORIGINAL
C(1,4)=B1
C(2,4)=B1
C(4,1)=B1
C(4,2)=B1
C(4,4)=A1

470 CONTINUE

PG=PF(IP1)
CALL PRSIS(STRESS,PG,GSIRES)
CALL MODCON(C,ntb,PG,GSIRES)

IF (IFLAG.NE.0) GO TO 630

      CALL ITRESLD(NEL,IP1,GSIRES,FACN,PS,TT,PG,LM,XP)
RETURN
630 CONTINUE

```

```

00300*17
00301*17
00302*14
00303*14
00304*14
00305*14
00306*14
00307*14
00308**5
00309*17
00310*17
00311*17
00312*12
00313*11
00314**5
00315**5
00316
00317
00318
00319**6
00320**6
00321
00322
00323
00324
00325
00326
00327**3
00328**3
00329**3
00330
00331
00332
00333
00334
00335
00336**3
00337**3
00338**4
00339
00340*10
00341*10
00342
00343**4
00344*12
00345*12
00346*12
00347*12
00348*12
00349*14
00350*14
00351*14
00352*17
00353*17
00354*14
00355*14
00356*14
00357*14
00358*14
00359*11

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IF (140) 809,700,809	00360*17
809 CONTINUE	00361*17
	00362*17
	00363*12
	00364*12
	00365**5
	00366**5
	00367**5
IF (1STRES.EQ.15IKKR) GO TO 700	00368**5
RETURN	00369
	00370
	00371
PRINTING OF STRESSES	00372
	00373
700 IF (IPED.EQ.1) GO TO 800	00374
	00375
DM = (STRESS(1)+STRESS(2)+STRESS(4))/3.	00376
DX = STRESS(1) - DM	00377
DY = STRESS(2) - DM	00378
DZ = STRESS(4) - DM	00379
SBAR=DSQRT(.5*(DX*DX+DY*DY+DZ*DZ))+DS*DS)	00380
FT=3.*ALFA*DM + SBAR - YLD	00381
	00382
	00383
800 CONTINUE	00384
IF (IPT-1) 810,808,810	00385**3
	00386
	00387
802 CONTINUE	00388**3
808 IF (ITYF20) 803,805,803	00389
803 WRITE(6,2002)	00390*12
GO TO 806	00391
805 WRITE(6,2003)	00392*12
	00393*12
806 CONTINUE	00394**3
WRITE (6,2004) NED	00395
	00396
	00397*17
	00398*17
810 CALL MAXMIN(STRESS, SX, SY, SM, DLRST, SHSTN, FMAX, PG, IPT, TIME, NUNE	00399*17
* L, SP, SD, SHM)	00400*19
	00401*22
	00402*19
IF (ITYF20) 813,815,813	00403
	00404
813 WRITE (6,2005) IPT, STATE(IPED), (STRESS(1), I=1,3), SX, SY, SM, FT, PF(IP	00405**9
*T)	00406**8
	00407**8
GO TO 817	00408*17
	00409
815 WRITE (6,2007) IPT, STATE(IPED), STRESS(4), (STRESS(1), I=1,3),	00410
1 SX, SY, SM, FT, PF(IPT)	00411**8
	00412*17
	00413*17
	00414*17
817 CONTINUE	00415*17
	00416*17
	00417*17
IF (NEL.FQ.0) GO TO 816	00418*17
LN=RELGMT(I)	00419*17

```

IF (IP.NE.NEL) GO TO 816
IF (IP.NE.1) GO TO 816
WRITE ON TAPE NO. 57 *****
SN=SP
IF (NFINCH.NE.0)
*WRITE(57)NEL,(STRESS(1),I=1,2),STRESS(4),STRESS(3),(STRAIN(1),I=1,
*2),STRAIN(3),STRESS(5),STRESS(6),STRESS(8),SM,SD,PG,SHM,EPSV
IP=IP+1
816 CONTINUE
***** CONTOUR PLOTTING COUNTER *****
IF (JPRINT.LT.1START) GO TO 819
IF (IP.NE.1) GO TO 819
WRITE INFORMATION ON TAPE NO. =18 ****88
IF (NFINCH.NE.0)
*WRITE(18)NEL,STRESS(5),STRESS(6),STRESS(8),SM,SD,
*PG,SHM,EPSV
819 CONTINUE
RETURN
2002 FORMAT (90H ELEMENT STRESS
1 ,21X,5HYIELD/99H NUM/IPT STAT
2E STRESS-YY STRESS-ZZ STRESS-YZ MAX STRESS
3IN STRESS ANGLE,9X,8HFUNCTION ,2X,13HPURE PRESSURE / )
2003 FORMAT (104H ELEMENT STRESS
1 ,21X,5HYIELD /
2 113H NUM/IPT STATE STRESS-XX STRESS-YY STRESS-ZZ
3 STRESS-YZ MAX STRESS MIN STRESS ANGLE,9X,8HFUNCTION
* ,2X,13HPURE PRESSURE / )
2004 FORMAT (14)
2005 FORMAT (5X,12,2X,A1,6HPLASTIC,1X,3E14.6,3X,2E14.6,3X,F6.2,3X,E14.6,
*2X,E12.5)
2007 FORMAT (5X,12,2X,A1,6HPLASTIC,1X,4E14.6,3X,2E14.6,3X,F6.2,3X,E14.6,
*2X,E12.5)
2100 FORMAT (1H1,21H STOP DRUCKER-PRAGER )
END

```

```

00420*17
00421*17
00422*17
00423*17
00424*17
00425*19
00426*22
00427*19
00428*20
00429*20
00430*19
00431*19
00432*17
00433*17
00434*17
00435*17
00436*17
00437*17
00438*17
00439*17
00440*17
00441*17
00442*17
00443*19
00444*20
00445*20
00446*20
00447*21
00448*17
00449*17
00450
00451
00452
00453
00454
00455
00456**8
00457**8
00458
00459
00460
00461
00462**8
00463**8
00464
00465**8
00466**8
00467**8
00468**8
00469
00470
00471

```


C SUBROUTINE FIGEN(A,R,H,MV)
E DIMENSION A(1), K(1)

C DOUBLE PRECISION A,K,ANORM,ANRMS,THR,X,Y,SINX,SINX2,COSX,
E COSX2,SINCS, RANGE
C 11 CONTINUE

C 5 RANGE = 1.0E-12
E IF (MV-1) 10,25,10
C 10 10 = -N

C DO 20 J=1,M
E 10 = 10 + N
C DO 20 I=1,M
E 10 = 10 + 1
C R(I) = 0.0
E IF (I-J) 20, 15, 20
C R(J) = 1.0
E 20 CONTINUE

C COMPUTE INITIAL AND FINAL NORMS

C 25 ANORM = 0.0
E DO 35 I=1,M
C DO 35 J=1,M

C 30 IA = 1 + (J-I) / 2
E ANORM = ANORM + A(IA) * A(IA)
C 35 CONTINUE

C 40 ANORM = 1.414 * DSORT(ANORM)
E ANRMS = ANORM * RANGE / FLOAT(N)

C INITIALIZE INDICATORS AND COMPUTE IMMEDIATES

C 45 IND = 0
E THR = ANORM
C 50 THR = THR / FLOAT(N)
E L = 1
C 55 M = L + 1

C COMPUTE SIN & COS
C 60 MU = (M*N-M) / 2
E LI = (L+L-1) / 2
C 65 DM = 1 + MU
E LI = L + LI
C 65 IND = 1
E MM = M + MU
C 68 X = 0.5 * (A(LI)-A(M))
E Y = -A(LI) / DSORT(A(LI) * A(LI) + X * X)
C 70 IF (X) 70,75,75
E 70 = 70

C 62 IF (DABS(A(LI))-THR) 130,65,65

C 65 IND = 1
E LI = L + LI
C 68 MM = M + MU
E X = 0.5 * (A(LI)-A(M))
C 70 Y = -A(LI) / DSORT(A(LI) * A(LI) + X * X)
E IF (X) 70,75,75

C 75 SINX = Y / DSORT(2.0 * (1.0 + (DSORT(1.0 - Y*Y))))
E SINX2 = SINX * SINX
C 78 COSX = DSORT(1.0 - SINX2)
E COSX2 = COSX * COSX

SINCS = SINX * CUBA

ROUTINE 1 AND M COLUMNS

ILQ = M*(L-1)

IMQ = M*(M-1)

DO 125 I=1,N

LU = (I+I-1) / 2

IF (I-L) 80,115,80

IF (I-M) 85,115,90

IM = I+M

GO TO 95

IM = M + LU

IF (I-L) 100,105,105

IL = I + LU

GO TO 110

IL = L + LU

X = A(IL) * CUSX - A(IM) * SINX

A(IM) = A(IL) * SINX + A(IM) * CUSX

K(IL) = X

IF (M-1) 120,125,120

ILK = ILQ + 1

IMK = IMQ + 1

X = K(ILK) * CUSX - K(IMK) * SINX

K(IMK) = K(ILK) * CUSX

K(ILK) = X

CONTINUE

X = A(LU) * SINCS

X = A(LU) * CUSX2 + A(M) * SINX2 - X

X = A(LU) * SINX2 + A(M) * CUSX2 + X

A(LM) = (A(IL) - A(M)) * SINCS + A(LM) * (CUSX2 - SINX2)

K(LM) = X

A(MN) = X

TESTS FOR COMPLETION

TEST FOR M = LAST COLUMN

IF (M-N) 135,140,135

GO TO 60

TEST FOR L = SECOND FROM LAST COLUMN

IF (L-(N-1)) 145,150,145

L = L + 1

GO TO 55

IF (M-1) 160,155,160

MD = 0

GO TO 50

COMPARE THRESHOLD WITH TRANS EIGENVECTOR

IF (P(K-AMK)) 165,165,45

SORT EIGENVALUES AND EIGENVECTORS

IF (I-M) 165,165,165

DO 185 I=1,N

GO TO 10

GO TO 10

GO TO 10

GO TO 10

GO TO 10

GO TO 10

GO TO 10

GO TO 10

GO TO 10

GO TO 10

GO TO 10

```

LL = I + (J+I-1)/2
JG = H + (I-2)
DO 185 J=1, H
MM = J + (J+J - J) / 2
WRITE(6,169) I, IO, LL, JG, J, JG, MM
IF (A(LL) - A(MM)) 170, 165, 185
170 X = A(LL)
A(LL) = A(MM)
A(MM) = X
WRITE(6,169) I, IO, LL, JG, J, JG, MM, X, A(LL), A(MM)
FORMAT(5X,115,3E16.3)
175 IF (MV - 1) 175, 185, 175
DO 180 K=1, N
ILK = IO + K
IKR = JG + K
X = K(ILK)
K(IKR) = K(ILK)
K(IKR) = X
CONTINUE
KFLURN
END

```

C
C 169
C
175
180
185

```

C      DATA SET ELEMENT      AT LEVEL 026 AS OF 83/02/18
SUBROUTINE ELEM1(MUMMAT)      00001**4
IMPLICIT REAL*8 (A-H,I-Z)    00002
                                00003
                                00004
                                00005
SUBROUTINE ELEM1 CALLS ELSSTIF WHICH FORMS ELEMENT STIFFNESS 00006
MASSES, AND DAMPING MATRICES 00007
                                00008
                                00009
COMMON/CORTEL/DT,RED(6),RADN,NUMNP,NUMEL,NEQ,MBAND,NCYCL,NDYN,IPLNA 00010
*A,ISINCH,NEPINT,ISTART,MECH,MBANDH,NEQS,MBANDS 00011**5
COMMON/QUAD/CCCC(1500),N1,N2,N3,N4,N5,MTOT,MNN,NK15 00012*20
                                00013*19
                                00014*19
COMMON/INTEGE/PETA,BETA,GAMA,KTOD,NGAMA,NEUNCH,NEGUB,ITEMX 00015*24
COMMON/BELA/NS,NELTYP,ND,NUMGT,MODEL,NCUN,LDW,NK015,NK016,NK017 00016*15
                                00017
                                00018*25
COMMON/BOHBY/KKS,NUMDT,NC,LOOP,NDRN,NINT,LDWA,NPT 00019*25
COMMON/CLOCK/TIM1,TIM2,TIM3 00020*25
COMMON D(1) 00021*23
                                00022*23
                                00023
REWIND 14 00024*17
                                00026**7
IF(NC.GT.0.AND.KKS.EQ.3) GO TO 500 00027**7
READ(5,1000) NUMMAT,NELTYP,NS 00028*11
WRITE(6,2000) NUMMAT,NELTYP,NS 00029*11
IF(KKS.EQ.3) WRITE(14) NUMMAT,NELTYP,NS 00030*11
GO TO 600 00031**7
                                00032**7
                                00033**7
500 READ(14) NUMMAT,NELTYP,NS 00034*11
600 CONTINUE 00035**7
                                00036
                                00037**7
                                00038**7
                                00039*11
IF(NS.EQ.0) NS=4 00040*11
IF(NELTYP.EQ.0) ND=8 00041*11
IF(NELTYP.EQ.1) ND=12 00042*11
IF(NELTYP.EQ.2) ND=16
                                00043**7
                                00044**7
                                00045**7
                                00046**7
                                00047**7
                                00048**7
                                00049**7
                                00050**7
                                00051*26
                                00052*12
***** CHANGE FOR ELASTO PLASTIC ANALYSIS ***** 00053*12
                                00054*12
                                00055*12

```

	NK015=NK14+NUMEL	00056*26
	NK016=NK015+NUMEL	00057*12
	NK017=NK016+NUMMAT	00058*12
	IDWA=100*NINT*NINT	00059*19
	NK15=NK017+IDWA*NUMEL	00060*19
	NPI=NINT*NINT	00061*19
C		00062**7
C	IF(NK15.LE.#T01) GO TO 20	00063**7
	NNN=0	00065
	GO TO 100	00066
C	20 CONTINUE	00067
		00068**7
C	WRITE(6,1) N1,N2,N3,N4,N5,NK6,NK7,NK8,NK9,NK10,NK11,NK12,NK13,NK14	
C	*,NK015,NK016,NK017,NK15,NUMNF,NUMMAT	
C		00069**7
C	CALL ELSTIF(D(N1),D(N2),D(N3),D(N4),D(N5),D(NK6),D(NK7),D(NK8),	00070*10
C	* D(NK9),D(NK10),D(NK11),NUMNF,NUMMAT,D(NK12),D(NK13),D(NK14),D(NK0	00071*11
C	*15),D(NK016),D(NK017))	00072*12
		00073*13
		00074*12
		00075*11
C	2 FORMAT(5X,2I5,10E10.3/15X,10E10.3/15X,10E10.3/15X,10E10.3/15X,10E	
	*10.3/)	
C	1 FORMAT(5X,2I5,2Z15)	
C	WRITE(6,4)	
C	4 FORMAT(5X,' PRINT IN ELEMENT')	
C	J=N3-1	
C		
C		
C	WRITE(6,2) N2,J,(D(I),I=N2,J)	
	J=N4-1	
C	WRITE(6,2) N3,J,(D(I),I=N3,J)	
	J=N5-1	
C	J=NK6-1	
C	WRITE(6,2) N5,J,(D(I),I=N5,J)	
	J=NK7-1	
C	WRITE(6,2) NK6,J,(D(I),I=NK6,J)	
		00076*21
C	J=NK8-1	
C	WRITE(6,2) NK7,J,(D(I),I=NK7,J)	
	J=NK9-1	
C	WRITE(6,2) NK8,J,(D(I),I=NK8,J)	
	J=NK10-1	
C	WRITE(6,2) NK9,J,(D(I),I=NK9,J)	
	J=NK11-1	
C	WRITE(6,2) NK10,J,(D(I),I=NK10,J)	
	J=NK12-1	
C	WRITE(6,2) NK11,J,(D(I),I=NK11,J)	
	J=NK13-1	
C	WRITE(6,2) NK12,J,(D(I),I=NK12,J)	
	J=NK14-1	
C	WRITE(6,2) NK13,J,(D(I),I=NK13,J)	
		00077*21

```

C J=NK015-1
  WRITE(6,2) NK14,J,(D(I),I=NK14,J)
C J=NK016-1

```

```

C J=NK017-1
  WRITE(6,2) NK016,J,(D(I),I=NK016,J)
C J=NK15-1
  WRITE(6,2)NK017,J,(D(I),I=NK017,J)
C

```

```
100 RETURN
```

```
00078
```

```
00079
```

```
00080*11
```

```
00081*11
```

```
00082*11
```

```
00083*11
```

```
00084*11
```

```
00085*11
```

```
00086*11
```

```
00087*11
```

```
00088*11
```

```
00089
```

```
1000 FORMAT(3I5)
```

```
2000 FORMAT(1H1/)
```

```

*      NUMBER OF MATERIALS = ',15/
*      CODE FOR ELEMENT TYPE = ',15/
*      EQ.0. 4-NODED RECTANGULAR ELEMENT (LINEAR)
*      EQ.1. 4-NODED RECTANGULAR ELEMENTS WITH INCOMPATIBLE MODES
*      EQ.2. 8-NODED RECTANGULAR ELEMENTS WITH QUADRATIC MODESS
*      NUMBER OF STRESS OUTPUT = ',15/
*      EQ.0. STRESS OUTPUT REQUESTED AT CENTER
*      EQ. 20 . STRESS OUTPUT REQUESTED AT CENTER PLUS MIDSIDE
END

```

DATA SET ELINI AT LEVEL 010 AS OF 82/12/02
SUBROUTINE ELINI(GMS,NE,X,Y,VOL,TTX0,TTX1,TTA2,NEL)

IMPLICIT REAL*8 (A-H,O-Z)

00001**7
00002**5
00003**5
00004
00005**5
00006**5
00007**8
00008**8
00009**4
00010**4
00011**4
00012**4
00013**9
00014**10
00015**3
00016**4
00017**4
00018**4
00019**4
00020**4
00021**4
00022**4
00023**4
00024
00025
00026
00027
00028
00029**4
00030**4
00031**4
00032**4
00033**4
00034**4
00035**4
00036**4
00037
00038
00039**5
00040**8
00041
00042
00043
00044
00045
00046
00047
00048**9
00049**9
00050**9
00051**9
00052
00053
00054
00055
00056
00057
00058
00059

COMMON/EMB/DB(16,16),GM(16),GH(16,16),BB(16,16),XP(16),XN(
*16),LB(16)
DIMENSION TT1(12),TT2(12),TTX1(2,12),TTX0(12,2),TTX2(12,12)
COMMON/QUAD/RR(4),ZZ(4),FAC,R,U(6),P(4,12),SF(12,12),H(12,12),
*SI(20,12),D(4,4),C(12,12),E(12,12),FF(5,12),PI(4,8)
COMMON/BFLA /NS,NELTYP ,ND

COMMON/RANI/RM(5),ZM(5),IP1
DIMENSION GMS(8),BK(8),DH(8),X(1),Y(1)

***** INITIALIZATION OF ELEMENT MATRICES *****

DO 85 I=1,ND
DO 85 J=1,ND
SF(I,J)=0.
H(I,J)=0.
F(I,J)=0.
C(I,J)=0.
85 CONTINUE

DO 860 I=1,ND
DO 860 J= 1,ND
ST(I,J)=0.0
860 CONTINUE
DO 90 J=1,8
GMS(J)=0.0
DO 80 I=1,16
GM(I)=0.
XP(I)=0.0
XN(I)=0.0
DO 80 J=1,16
S(I,J)=0.0
SS(I,J)=0.0
GH(I,J)=0.0
80 CONTINUE

DO 100 I=1,4
RM(I)=0.0
ZM(I)=0.0
100 CONTINUE

CALCULATES VOLUME HERE FOR MASS CALCULATION CHANGE HERE FD00058
SISTENT MASS MATRICES 00059

***** IN IT ALSO U.S. FOR INJANGLE ? *****

```
R21=RR(2)-RR(1)
Z31=ZZ(3)-ZZ(1)
R31=RR(3)-RR(1)
Z41=ZZ(4)-ZZ(1)
Z21=ZZ(2)-ZZ(1)
R41=RR(4)-RR(1)
VOL=(R21*Z31 +R31*Z41-R31*Z21-R41*Z31)
VOL=VOL./2.
```

```
IF(VOL.GT.0.0) GO TO 61
WRITE(6,2012) VOL
STOP
81 CONTINUE
```

```
IF(IJUMP.NE.0) GO TO 615
DO 8615 J=1,2
  DO 8615 J=1,ND
    TTX0(J,J)=0.0
    TTX1(I,J)=0.0
  DO 8616 I=1,ND
    DO 8616 J=1,ND
      TTX2(I,J)=0.0
```

615 CONTINUE

2012 FORMAT (//) ERROR IN ELEMENT CARD = ',110/') END

00060**6
00061**6
00062**6
00063
00064
00065
00066
00067
00068
00069**2
00070
00071
00072
00073
00074
00075
00076
00077
00078
00079
00080
00081
00082
00083**4
00084**4
00085**4
00086
00087
00088**4
00089
00090
00091**4
00092**4
00093
00094
00095**2
00096
00097
00098


```

DATA SET EDNIF      AI LEVEL 027 AS OF 82/06/12
SUBROUTINE EDNIF(FK, DBETA1, DBETA2, NUMMAT, ELMDA, EMU, EALFA, EM, EK, ROS00001
00002*13
*, ROF, MBAND, MBANDH, NEL, ANK, INEL, IMP, IMAT, KN, R0UMF, IJUMP, IKS, IKF,
00003
00004*13
*PRESS, X, Y, RH, ZH, RR, ZZ, MD, NUMEL, NP, MBANDS, THICK)
00005**7
00006*13
IMPLICIT REAL*8 (A-H, O-Z)
00007
00008**9
00009**9
00010**9
COMMON/EDNIF/RS, NUMGT, NC, LUMP, NDRN, NINT, IDWA, NPT
00011*14
DIMENSION INP(4), NP(4), ELMDA(1), EMU(1), EALFA(1), EM(1), EK(1),
00012
00013*13
*ROS(1), ROF(1), X(1), Y(1), RR(4), ZZ(4)
00014**8
00015
DIMENSION FK(1), DBETA1(1), DBETA2(1)
00016**7
DIMENSION PRESS(5)
00017*11
00018*25
DIMENSION FM(5), ZM(5)
00019*27
00020
00021*11
00022*11
IF(MD.GT.0) GO TO 40
00023
IF(NEL.NE.0) GO TO 800
00024
READ ELEMENT INFORMATION AND CALL LOAD ROUTINE FOR CALCULA
00025
LOAD VECTOR FOR STATIC ANALYSIS
00026
00027
00028
00029*21
00030*21
00031*21
00032*20
00033
00034**9
00035**9
00036*15
00037*15
***** BRANCH HERE FOR ELASTOPLASTIC ANALYSIS *****
00038*15
00039*15
00040*20
00041*20
00042*20
00043
DO 20 N=1, NUMMAT
00044**9
READ(5, 1003) ELMDA(N), EMU(N), EALFA(N), EM(N), EK(N), ROS(N), ROF(N),
00045**6
*FK(N), DBETA1(N), DBETA2(N)
00046**6
00047**9
00048*16
WRITE(6, 2003) ELMDA(N), EMU(N), EALFA(N), EM(N), EK(N), ROS(N), ROF(N)
00049*17
*, FK(N), DBETA1(N), DBETA2(N)
00050*17
00051**9
00052*15
00053*15
00054*15
00055**9
00056
00057
00058**6
00059
100 CONTINUE
20 CONTINUE
MBAND=0
MBANDH=0
MBANDS=0
NEL=0

```

```

XKK=0.577350269169626          00060
IF (NINT.EQ.1) XKK=0.0
IF (NC.GI.U.AND.KKS.EQ.3) GO TO 800 00061*24
WRITE(6,2006)                   00062
800 CONTINUE                     00063
IF (NMEL=NEL) 50,700,30        00064
                                00065**9
                                00066**9
                                00067**9
30 CONTINUE                     00068**9
IF (NC.GI.U.AND.KKS.EQ.3) GO TO 200 00069**9
                                00070**9
READ(5,1007) INEL,INP,IMAT,KN,KJUMP,IKS,IKF,PRESS,THICK 00071**9
IF (KKS.EQ.3) WRITE(14) INEL,INP,IMAT,KN,KJUMP,IKS,IKF,PRESS, 00072**9
*THICK                          00072**9
GO TO 300                       00073**9
200 READ(14) INEL,INP,IMAT,KN,KJUMP,IKS,IKF,PRESS,THICK 00074**9
300 CONTINUE                    00075**9
                                00076**9
                                00077**9
                                00078**9
                                00079
40 IJUMP=0                      00080
NEL=NEL+1                       00081
ML=INEL-NEL                     00081
IF (ML) 50,55,00               00082
50 WRITE(6,2007) INEL         00083
STOP                             00084
                                00085**9
                                00086**9
                                00087**9
                                00088
55 DO 56 I=1,4                 00089
56 NP(I)=INP(I)                00090
MAT=IMAT                       00091
KS=IKS                         00092
KF=IKF                         00093
GO TO 62                       00094
60 DO 61 I=1,4                 00094
61 NP(I)=NP(I)+KN              00095
IF (KJUMP.NE.0) IJUMP=1        00096
62 CONTINUE                    00097*23
IF (NC.GI.U.AND.KKS.EQ.3) GO TO 63 00098*23
                                00099*23
WRITE(6,2006) NEL,NP,MAT,KS,KF,PRESS,THICK 00100*23
WRITE(49,1007) NEL,NP,MAT,KN,KJUMP,KS,KF,PRESS,THICK 00100*23
                                00101*23
63 CONTINUE                    00102*23
                                00103**9
                                00104**9
***** CHECK ALSO FOR TRIANGLE CENTROID CALCULATION ARBIT 00105*14
                                00106*14
DO 70 I=1,4                   00107
M=NP(I)                       00108
RR(I)=X(M)                    00109
70 ZZ(I)=Y(M)                 00110
                                00111*25
                                00112*25
RR(5)=0.25*(RR(1)+RR(2)+RR(3)+RR(4)) 00113*25
ZZ(5)=0.25*(ZZ(1)+ZZ(2)+ZZ(3)+ZZ(4)) 00114*25
                                00115*25
700 CONTINUE                   00116

```

RETURN	00117
	00118**9
	00119**9
1003 FORMAT(0F10.3)	00120**6
1007 FORMAT(5I3,5I2,5E10.3,F5.2)	00121*12
2002 FORMAT (//' * 7X,'E',11X,'NU',10X,'N',11X,'M',11X,'K',10X,'RDS',9X,'ROF',5X,'FK' * 5X,'DBETA1',5X,'DBETA2'//)	00122 00123**6 00124**7
2003 FORMAT(5X,10E12.3)	00125**6
2006 FORMAT (10I1' *25X,'MATERIAL'/3X,'NUMBER',10X,'1',9X,'2',9X,'3',9X,'4',11X,'NUMBE' *R',40X,'ELEMENT DATA'//3X,'ELEMENT',14X,'NODE' NUMBERS' *R',40X,'ELEMENT CARD ERROR, ELEMANT'//)	00126 00127 00128
2007 FORMAT (//' 2008 FORMAT(5X,15,4I10,5X,110,2I3,5E10.3,3X,F5.2)	00129 00130*19
END	00131

```

DATA SET IELPAL, AT LEVEL 042 AS OF 03/03/05
SUBROUTINE IELPAL(PROP,SIG,EPS,YIELD,IPEL,PF,PS,FACN,LM,PMAX,
*NUMBER)
IMPLICIT REAL*8(A-H,O-Z)
.....
:
: ISI NUMBER OF STRESS COMPONENTS :
: ISR NUMBER OF STRAIN COMPONENTS :
SIG STRESSES AT THE END OF THE PREVIOUS UPDATE :
EPS STRAINS AT THE END OF THE PREVIOUS UPDATE :
RATIO PART OF STRAIN INCREMENT TAKEN ELASTICALLY :
DELEPS INCREMENT IN STRAINS :
DELSIG INCREMENT IN STRESSES, ASSUMING ELASTIC BEHAVIOR :
PROP(1) YOUNG S MODULUS :
PROP(2) POISSON S RATIO :
PROP(3) INITIAL YIELD STRESS IN SIMPLE TENSION :
PROP(4) STRAIN HARDENING MODULUS :
IPEL = 1, MATERIAL ELASTIC :
: = 2, MATERIAL PLASTIC :
:
:
COMMON/EL/END,ICOUNT
COMMON/VAR/NG,KPRI
COMMON /VMISEB/ A1,D1,C1,D1,A2,D2,C2,D2,YLD,SM,ISR,IST
COMMON/HATMOD/STRESS(9), STRAIN(4),C(4,4),IPT,NEL
DIMENSION PROP(NCON),SIG(4),EPS(4)
DIMENSION IAW(4),DELSIG(4),DELEPS(4),DEPS(4),STATE(2)
COMMON/SURIN/TT(5,12),XP(16),INDNL,ITYP2D
COMMON/IDIEGE/TETA,BETA,GAMA,RTOL,NGAMA,NPUNCH,NEQUIB,IEMX
COMMON/BOEBY/KKS,NUMMAT,NC,LOOP,NDRN,HINT,IDWA,NPT
COMMON/DELA/PS,NEDEIP,NO,NUMGT,MODEL,NEON,IPW,NK015,NK016,NK017
COMMON/CUNTL/DD,HED(8),RADN,NUMNP,NUMSL,NEG,MBAND,NCYCL,NDYN,IPLNA
*X,ISTPR,NPRINT,ISTART,NEQH,MBANDH
COMMON/PRIN/ISTRES,MPRINT,OPRINT,KRINT,IP
COMMON/PUBA/ALFA,EM,PPSV,SHMD
COMMON/KALI/JPT,KEL,JPLNT(200),KELMT(200)
COMMON/CHURDA/WP
DIMENSION PMAX(NUMEL,1)
DIMENSION PF(1)
DIMENSION GSTRES(9)
DIMENSION PS(16,8),FACN(4),LM(16)
EQUIVALENCE (DELEPS(4),DEPS(4))

```

```

00001*32
00002*28
00003*28
00004*28
00005
00006
00007
00008
00009
00010
00011
00012
00013
00014
00015
00016
00017
00018
00019
00020
00021
00022
00023
00024
00025
00026
00027
00028
00029**2
00030**2
00031
00032*29
00033
00034**7
00035
00036*28
00037*27
00038
00039*33
00040*33
00041*33
00042**4
00043**4
00044**7
00045**7
00046**7
00047*28
00048*42
00049*28
00050*28
00051*28
00052**7
00053*10
00054*31
00055*25
00056**7
00057**5
00058

```

```

COMMON/GORA/IFLAG, IIR
DATA NGHAST/1000/, STATE/INE, IIR/
*P=0.0
TIME=IND*DD
IF (IPT.NE.1) GO TO 110
IST=4
IF (ITYP2D.EQ.2) IST=3
ISR=3
IF (ITYP2D.EQ.0) ISR=4
YM=PROP(1)
PV=PROP(2)
D1=PV/(PV - 1.)
A2=YM/(1.+PV)
B2=(1.-PV)/(1.-2.*PV)
C2=PV/(1.-2.*PV)
D2=YM*PROP(4)/(YM-PROP(4))
C1=A2/2.
B1=YM/(1. - 2.*PV)/3.
***** UNDRAINED CHARACTERISTIC *****
IF (ITYP2D.EQ.2) GO TO 105
PLANE STRAIN / AXISYMMETRIC
B1=A2*C2
A1=B1+A2
GO TO 110
PLANE STRESS
105 A1=YM/(1.-PV*PV)
B1=A1*PV
110 YLD = YIFLD
1. CALCULATE INCREMENTAL STRAINS
DO 120 I=1,ISR
120 DELEPS(I) = STRAIN(I) - EPS(I)
2. CALCULATE THE STRESS INCREMENT,
ASSUMING ELASTIC BEHAVIOR
DELSIG(1) = A1*DELEPS(1) + B1*DELEPS(2)
DELSIG(2) = B1*DELEPS(1) + A1*DELEPS(2)
DELSIG(3) = C1*DELEPS(3)
DELSIG(4) = 0.
IF (ITYP2D.EQ.2) GO TO 150

```

```

00059*24
00060*24
00061*30
00062
00063*33
00064*33
00065*33
00066*33
00067
00068*28
00069
00070
00071
00072
00073
00074
00075
00076
00077
00078
00079
00080
00081
00082
00083
00084
00085
00086
00087*28
00088*28
00089*28
00090
00091
00092
00093
00094
00095
00096
00097
00098
00099
00100
00101
00102
00103
00104
00105
00106
00107
00108
00109
00110
00111
00112
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DELSIG(1) = D1 * (DELEPS(1)+DELEPS(2))          00118
IF (IYF2D.EQ.1) GO TO 150                       00119
DELSIG(1) = DELSIG(1) + D1*DELEPS(4)          00120
DELSIG(2) = DELSIG(2) + D1*DELEPS(4)          00121
DELSIG(4) = DELSIG(4) + A1*DELEPS(4)          00122
                                                00123
                                                00124
3. CALCULATE TOTAL STRESSES,                     00125
   ASSUMING ELASTIC BEHAVIOR                     00126
150 TAU(4) = 0.                                  00127
DO 160 I=1,151                                   00128
160 TAU(I) = SIG(I) + DELSIG(I)                 00129
                                                00130
4. CHECK WHETHER *TAU* STATE OF STRESS FALLS   00131
   OUTSIDE THE LOADING SURFACE                 00132
SM = (TAU(1)+TAU(2)+TAU(4))/3.                 00133
SX = TAU(1) - SM                               00134
SY = TAU(2) - SM                               00135
SZ = TAU(4) - SM                               00136
SI = TAU(1) - SM                               00137
SS = TAU(2) - SM                               00138
SZ = TAU(4) - SM                               00139
                                                00140
F1 = .5*(SX*SX+SY*SY+SZ*SZ) + SS*SS - YLD*YLD/3. 00141
***** TEMPORARY PRINT OUT *****            00142*30
                                                00143*39
161 WRITE(6,161)                                00144*39
   FORMAT(SX, ' *** PRINT IN EDPAL. ***',/)    00145*39
162 WRITE(6,162) NC, ITR, IFLAG, NEL, IPT      00146*30
   FORMAT(SX,1615)                             00147*30
   WRITE(6,163) (STRAIN(I), I=1,4)            00148*30
   WRITE(6,163) (EPS(I), I=1,4)              00149
   WRITE(6,163) (DELEPS(I), I=1,4)          00150
   WRITE(6,163) (SIG(I), I=1,4)             00151
   WRITE(6,163) (DELSIG(I), I=1,4)         00152
   WRITE(6,163) (TAU(I), I=1,4)            00153
   WRITE(6,163) (STRESS(I), I=1,4)        00154
   WRITE(6,163) SM, SX, SY, SS, SZ, YLD, PF(IPT), FT 00155
163 FORMAT(SX,1610,3)                          00156
                                                00157
IF (F1) 170,170,300                             00158
STATE OF STRESS WITHIN LOADING SURFACE - ELASTIC BEHAVIOR 00159
170 IPEL=1                                       00160
STRESS(4) = 0.                                  00161
DO 180 I=1,151                                   00162
180 STRESS(I) = TAU(I)                          00163
IF (ITYP2D.EQ.2) STRAIN(4)=EPS(4) + D1*(DELEPS(1) + DELEPS(2)) 00164
GO TO 400                                       00165
STATE OF STRESS OUTSIDE LOADING SURFACE - PLASTIC BEHAVIOR 00166

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300 IF (IPEL.EQ.1) GO TO 320                                00164
' .....WAS PLASTIC                                        00165
IPEL=2                                                    00166
RATIO = 0.                                                00167
DO 315 I=1,151                                           00168
315 TAU(I) = SIG(I)                                       00169
GO TO 370                                                 00170
' .....WAS ELASTIC                                        00171
DETERMINE PART OF STRAIN TAKEN ELASTICLY                 00172
320 IPEL=2                                                00173
SM = (SIG(1)+SIG(2)+SIG(4))/3.                            00174
SX = SIG(1) - SM                                         00175
SY = SIG(2) - SM                                         00176
SZ = SIG(4) - SM                                         00177
DM = (DELSIG(1)+DELSIG(2)+DELSIG(4))/3.                 00178
DX = DELSIG(1) - DM                                      00179
DY = DELSIG(2) - DM                                      00180
DS = DELSIG(3) - DM                                      00181
DZ = DELSIG(4) - DM                                      00182
A = DX*DX + DY*DY + 2.*DS*DS + DZ*DZ                    00183
B = SX*DX + SY*DY + 2.*SS*DS + SZ*DZ                    00184
E = SX*SX + SY*SY + 2.*SS*SS + SZ*SZ - 2.*YLD*YLD/3.   00185
RATIO=(-B+DSQR(B*B-A*E))/A                                00186
DO 350 I=1,151                                           00187
350 TAU(I) = SIG(I) + RATIO*DELSIG(I)                    00188
IF (ITYP2D.EQ.2) STRAIN(4)=EPS(4) + RATIO*D1*(DELEPS(1)  00189
+ DELEPS(2))                                             00190
*TAU NOW CONTAINS (PREVIOUS STRESSES +                 00191
STRESSES DUE TO ELASTIC STRAIN INCREMENTS)             00192
5. CALCULATE PLASTIC STRESSES                             00193
DETERMINE INCREMENT INTERVAL                             00194
370 M=20.*DSQR(F1)/YLD+1                                  00195
IF (M.(61.50) M=50                                       00196
XM = (1. - RATIO)/M                                      00197
DO 380 I=1,151                                           00198
380 DEPS(I) = XM*DELEPS(I)                                00199
' ..... CALCULATION OF ELASTOPLASTIC STRESSES ..... (START) 00200
DO 600 IM=1,M                                            00201
CALL MDPY (TAU,DEPS,C)                                    00202
00203
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DU 560 I=1,151
DU 560 J=1,15K
560 TAU(1) = TAU(1) + C(1,J) + DELT(D)

CORRECTION

DM = (TAU(1)+TAU(2)+TAU(4))/3.
DX = TAU(1) - DM
DY = TAU(2) - DM
DS = TAU(3)
DZ = TAU(4) - DM

IF (PROP(4).EQ.0.) GO TO 580

STRAIN-HARDENING MATERIAL - UPDATE YLD
YLD=DSQRT (1.5*(DX*DX+DY*DY+2.*DS*DS+DZ*DZ))
GO TO 600

PERFECTLY PLASTIC MATERIAL
580 F1A=.5*(DX*DX + DY*DY + DZ*DZ) + DS*DS
FTB=(YLD*YLD)/3.
F1=FTA - FTB
IF (F1.EQ.0) GO TO 600
IF (111F2D.EQ.2) GO TO 590

COEF=-1. +DSQRT(FTB/F1A)
TAU(1) = TAU(1) + COEF*DX
TAU(2) = TAU(2) + COEF*DY
TAU(3) = TAU(3) + COEF*DS
TAU(4)=TAU(4) + COEF*DZ
GO TO 600

590 COEF=DSQRT(FTB/F1A)
TAU(1)=TAU(1)*COEF
TAU(2)=TAU(2)*COEF
TAU(3)=TAU(3)*COEF
STRAIN(4)=STRAIN(4) + (COEF - 1.)*DM/DM

600 CONTINUE

..... CALCULATION OF ELASTOPLASTIC STRESSES ..... ( END )

STRESS(4) = 0.
DU 390 I=1,151
390 STRESS(1) = TAU(1)

400 CONTINUE

IF(IFLAG.EQ.0) GO TO 610

6. UPDATING STRESSES, STRAINS, YIELD, NS

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00226
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00252
00253
00254
00255
00256
00257
00258
00259
00260
00261
00262
00263
00264
00265
00266
00267*24
00268*24
00269
00270
00271
00272
00273
00274
00275*26
00276*32
00277*32
00278*32
00279*26
00280
00281
00282
00283

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DO 410 I=1,1ST
410 SIG(I) = STRESS(I)
DO 420 I=1,1SR
420 EPS(I) = STRAIN(I)
YIELD = YLD
IF (IYFZB.LM.2) EPS(4)=STRAIN(4)
610 CONTINUE

7. FORM THE MATERIAL LAW
IF (IPEL.EQ.1) GO TO 450
ELASTO-PLASTIC
CALL MIDFP (TAU,DEPS,C)

***** REALIGN THE CONSTITUTIVE MATRIX TO BE COMPATIBLE
***** CUMSV SUBROUTINE. ALSO CALCULATE THE RESIDUAL LOAD
***** OF ELEMENT LEVEL AND WRITE IT ON A TAPE TO BE ASSEMBLED
*** SUBROUTINE KSTAR.

PG=PF(IPT)
CALL PRINT(STRESS,PG,GSTRES)
CALL MIDCON(C,NEL,PG,GSTRES)

IF (NEQUIB.EQ.1.AND.IFLAG.NE.0) GO TO 620

CALL ITRSID(NEL,IPT,GSTRES,FACN,PS,TI,PG,LM,XP)

WRITE(6,165)
165 FORMAT(5X,' PRINT IN ELFAL ')
WRITE(6,166) ((C(I,J),J=1,4),I=1,4)
166 FORMAT(5X,'LE10.3')

WRITE(6,162) IFLAG,IPFL
WRITE(6,163) (GSTRES(I),I=1,4),FACN(IPT)
WRITE(6,163) (XP(I),I=1,16)
WRITE(6,162) (LM(I),I=1,16)

IF (NEQUIB.EQ.0) GO TO 620
RETURN
620 CONTINUE

IF(LMD) 820,700,820

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00284*32
00285*32
00286*32
00287
00288
00289
00290
00291
00292*24
00293*24
00294*10
00295*10
00296
00297
00298
00299
00300
00301
00302
00303
00304
00305**6
00306**6
00307**6
00308**6
00309**6
00310**6
00311**6
00312**6
00313**6
00314**6
00315*15
00316*15
00317*22
00318*22
00319*23
00320*22
00321*24
00322*24
00323*24
00324*24
00325*24
00326*28
00327*28
00328*24
00329*24
00330*24
00331*24
00332*24
00333*28
00334*28

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820	CONTINUE	00335*28	
		00336*28	
		00337*28	
		00338*24	
	IF (ISTRES.EQ.1STRPK) GO TO 700	00339*12	
	RETURN	00340**6	
		00341	
		00342	
	ELASTIC	00343	
450	DU 460 I=1,1SI	00344*13	
	DU 460 J=1,1SI	00345*13	
460	C(1,J)=0.	00346	
	C(1,1)=A1	00347	
	C(2,1)=B1	00348	
	C(1,2)=B1	00349	
	C(2,2)=A1	00350	
	C(3,3)=C1	00351	
	CHECK HERE FROM ORIGINAL *****	00352*10	
		00353*10	
440	CONTINUE	00354**9	
	C(1,4)=B1	00355	
	C(2,4)=B1	00356	
	C(4,1)=B1	00357	
	C(4,2)=B1	00358	
430	CONTINUE	00359	
	PG=PF(1P1)	00360**8	
	WRITE(6,166) A1,B1,C1	00361*22	
		00362*22	
		00363*22	
	CALL PRSIS(ISTRES,PG,GSTRES)	00364*23	
	CALL MODCON(C,NEL,PG,GSTRES)	00365*22	
		00366*26	
		00367*26	
	IF (NEQDID.NE.0.AND.IFLAG.NE.0) GO TO 630	00368*25	
		00369*25	
		00370*25	
		00371*28	
	CALL IFRSD(NEL,1P1,GSTRES,FACN,PS,TT,PG,LM,XP)		
	WRITE(6,165)		
165	FORMAT(5X,' PRINT IN ELA1')		
	WRITE(6,166) ((C(I,J),J=1,4),I=1,4)		
166	FORMAT(5X,4E10.3)	00372*25	
	WRITE(6,162) IFLAG,IPFL		
	WRITE(6,163) (GSTRES(I),I=1,4),FACN(1P1)		
	WRITE(6,163) (XP(I),I=1,16)		
	WRITE(6,162) (LM(I),I=1,16)		
	IF (NEQUIR.FQ.0) GO TO 630		
	RETURN	00373*25	
630	CONTINUE	00374*25	
		00375*25	
		00376*22	
		00377*13	
		00378*12	
	IF (IND) 809,700,809	00379*28	
		00380*28	

809	CONTINUE	00381*28
	IF(ISTRES.EQ.1STRPR) GO TO 700	00382*28
		00383*12
	RETURN	00384
		00385
		00386
		00387
	PRINTING OF STRESSES	00388
700	IF (IPEL.EQ.1) GO TO 705	00389
		00390
	DM=(STRESS(1) + STRESS(2) + STRESS(4))/3.	00391
	DX=STRESS(1) - DM	00392
	DY=STRESS(2) - DM	00393
	DS=STRESS(3)	00394
	DZ=STRESS(4) - DM	00395
	FT=.5*(DX*DX + DY*DY + DZ*DZ) + DS*DS - YLD*YLD/3.	00396
		00397
705	CONTINUE	00398
		00399**8
	SHM=SHMD	00400*18
	IF (IPT-1) 810,800,810	00401*42
		00402*18
800	CONTINUE	00403**8
		00404*10
802	CONTINUE	00405*10
		00406*10
	IF(NPUNCH.NE.0) GO TO 810	
808	IF (ITYP2D) 803,805,803	00407
803	WRITE(6,2002)	00408*22
807	CONTINUE	00409*16
	GO TO 806	00410
805	WRITE(6,2003)	00411*22
		00412*22
		00413
806	CONTINUE	00414*10
		00415*10
		00416*10
		00417*10
		00418*10
	WRITE (6,2004) NEL	00419
810	CALL MAXMIN(STRESS,SX,SY,SM,HLAST,SHSTN,PMAX,PG,IPT,TIME,NUME	00420*28
		00421*35
	* L,SP,SD,SHM)	00422*38
	IF(NPUNCH.NE.0) GO TO 817	
	IF (ITYP2D) 813,815,813	00423
		00424
813	WRITE (6,2005) IPT,STATE(IPEL),	00425
		00426*20
1	(STRESS(1),I=1,3),SX,SY,WP,FT ,PF(IPT)	00427*20
		00428*28
	GO TO 817	00429*28
		00430*28
		00431*28
		00432
815	WRITE (6,2007) IPT,STATE(IPEL),STRESS(4),	00433
1	(STRESS(1),I=1,3),SX,SY,WP,FT ,PF(IPT)	00434*20

817	CONTINUE	00435*28
		00436*28
		00437*28
		00438*28
		00439*28
	IF (KEL.FU,0) GO TO 816	00440*28
	LN=KEL.M1(IP)	00441*28
	IF (LN.NE.NEL) GO TO 816	00442*28
	IF (IPT.NE.1) GO TO 816	00443*28
		00444*28
	WRITE (ON TAPE NO. 57) *****	00445*28
		00446*35
		00447*38
	S=SP	00448*38
	IF (NPUNCH.NE.0)	00449*35
	*WRITE (57,2032)	00450*36
	*, NEL, (STRESS(1), I=1,2), STRESS(3), (STRAIN(1), I=1,	00450*36
	*2), STRAIN(3), STRESS(5), STRESS(6), STRESS(8), SM, SD, PG, SHM, EPSV, FT	00451*36
2032	FORMAT(13,7E11.4/8E10.3/)	
		00452*28
	IP=IP+1	00453*28
816	CONTINUE	00454*28
		00455*28
		00456*28
	IF (NC) 821,822,821	00457*40
		00458*40
		00459*40
		00460*40
821	CONTINUE	00461*40
	*****8 CONTOUR PLOTTING COUNTER *****8	00462*28
		00463*28
	IF (OPRINT.EI.18) GO TO 819	00464*28
822	CONTINUE	00465*40
		00466*40
		00467*40
		00468*40
	IF (IPT.NE.1) GO TO 819	00469*28
		00470*28
	WRITE INFORMATION ON TAPE NO. =18 *****	00471*28
		00472*35
		00473*35
	IF (NPUNCH.NE.0)	00474*35
	*WRITE (18,2033) NEL, STRESS(1), STRESS(2), STRESS(3), STRESS(8),	00475*36
	*PG, SHM, EPSV, STRAIN(3)	00476*37
2033	FORMAT(15,8E9.2,3X)	
		00477*37
		00478*35
		00479*28
819	CONTINUE	00480*28
		00481*28
	RETURN	00482
		00483
		00484
2002	FORMAT (90H ELEMENT STRESS STRESS-YY STRESS-ZZ STRESS-XX	00485
	1S-YZ MAX STRESS MIN STRESS,21X,5HYIELD/99H NUM/IPT STATO	00486
	2E	00487
	3 ANGLE,9X,8HFUNCTION ,2X,13HPURE PRESSURE /)	00488*20
2003	FORMAT (104H ELEMENT STRESS STRESS-XX STRESS-YY STRESS-YY	00489
	1ESS-ZZ STRESS-YZ MAX STRESS MIN STRESS,21X,5HYIELD /	00490
	299H NUM/IPT STAT	00491

3
 *2A, 13HQUE PRESSURE /)
 2005 FORMAT (5X, 12, 2X, A1, 6hLASTIC, 1X, 3E14.6, 3X, 2E14.6, 3X, E9.2, 1X, E14.6, 00492
 +1X, E12.5) 00493*20
 2007 FORMAT (5X, 12, 2X, A1, 6hLASTIC, 1X, 4E10.3, 3X, 2E14.6, 3X, E9.2, 1X, E14.6, 00494*20
 *1X, E12.5) 00495*20
 2004 FORMAT (14/) 00496*20
 00497*20
 00498
 00499
 00500

END

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C      DATA OF ELSVIF      AT LEVEL 0/1 AS OF 83/03/05
C      SUBROUTINE ELSVIF (ID,X,Y,IE,ELMDA,EMU,EALFA,EM,EK,RUS,ROP,NUMNP, 00001
C      *DBETA,FK,DBETA1,DBETA2,MATP,PRP,WA) 00002*30
C      IMPLICIT REAL*8 (A-H,O-Z) 00003*30
C      00004*31
C      00005
C      00006
C      00007
C      SUBROUTINE ELSVIF GENERATES MASSMATRICES(BOTH SOLID AND FLUID) 00008
C      MATRICES(BOTH FOR SOLID AND FLUID),AND STIFFNESS MATRICES(SOLID 00009
C      AND COUPLING MATRICES), LOAD VECTOR ELEMENTWISE FOR STATIC ANA 00010
C      PLANE STRAIN AND AXISYMMETRIC CASE) 00011
C      00012
C      00013
C      00014
C      COMMON/CONTI/DT,NEB(6),RADN,NASIM,NUMEL,NEQ,MBAND,NCYCL,NDYN,IFLNA 00015
C      *X,ISTRPR,NPRINI,ISTART,NEQH,MBANDH,NEQS,MBANDS 00016*12
C      COMMON/EMBS/S(16,16),GM(16),GH(16,16),SS(16,16),XP(16),XN( 00017*32
C      *16),LM(16) 00018*62
C      COMMON/INTEGE/PETA,BETA,GAMA,KIOL,NGAMA,NPUNCH,NEQUIB,ITEMX 00019*62
C      00020*64
C      00021*65
C      00022*50
C      COMMON/HUBBY/KKS,NUMGAI,NC,LODF,NDRN,NINT,IDWA,NPT 00023*58
C      COMMON/ANBN/FA 00024*20
C      COMMON/QUAD/RK(4),ZZ(4),FAC,K,U(6),P(4,12),SF(12,12),H(12,12), 00025*27
C      *ST(20,12),D(4,4),C(12,12),E(12,12),TF(5,12),P1(4,8),UDD(4,4) 00026*37
C      *,PS(16,8),FACN(4) 00027*63
C      00028*50
C      00029*29
C      00030*29
C      00031*29
C      00032
C      00033
C      00034
C      COMMON/ADHR/C1,C2,C3,AL,ALT,FACC,FACI 00035
C      00036
C      00037*57
C      00038*68
C      00039*57
C      COMMON/DIMEI/N43 00040*60
C      00041*60
C      00042
C      DIMENSION ID(NUMNP,1),X(1),Y(1),ELMDA(1),EMU(1),EALFA(1),EM(1), 00043
C      *EK(1),RUS(1),ROP(1),NF(4),IME(4),I11(2),SSS(2),IS(8),IL(8),IH(NUMNP 00044*12
C      *P,1),FK(1),DBETA1(1),DBETA2(1) 00045
C      00046
C      DIMENSION T11(12),T12(12),T1A1(2,12),T1X1(12,2),T1X2(12,12) 00047*27
C      DIMENSION GMS(8),LK(8),LH(8),PRESS(5) 00048*27
C      00049*29
C      00050*29
C      00051*29
C      DIMENSION PRP(NCON,1),WA(IDWA,1) 00052*57
C      00053*57
C      00054*57
C      00055*59
C      00056*59
C      DIMENSION MATP(1) 00057**2
C      ***** CALL ELEMENT INFORMATION ***** 00058**2
C      REWIND 11 00059*12

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REWIND 13 00060*12
REWIND 50 00061*12
REWIND 51 00062*12
REWIND 17 00063*12
          00064*69
          00065*31
          00066*46
42 FORMAT(5A,15) 00067*31
***** TAPE 11 CONTAINS ELEMENT INFORMATION FOR STRE00067*31
***** TAPE 13 CONTAINS ELEMENT MATRICES FOR AS00068*31
***** TAPE 50 AND 51 CONTAINS CONSTITUTIVE MATRI00069*31
STRAIN DISPLACEMENT MATRICES FOR SHAKEDOWN ANALYSIS **00070*31
          00071*31
          00072*31
          00073*12
          00074*12
          00075**8
          00076**8
40 CONTINUE 00077*3
          00078*32
          00079**6
          00080*29
          00081**2
          00082*29
          00083*13
          00084**2
          00085**2
          00086**2
          00087**2
          00088**2
          00089*61
          00090**2
          00091**2
          00092*42
          00093*42
          00094*42
          00095*42
          00096
          00097*50
          00098*50
          00099*50
          00100*50
          00101*35
          00102*37
          00103*37
          00104
          00105
          00106
          00107
4200 CONTINUE 00108
          00109
          00110
          00111*29
          00112
          00113
          00114*12
          00115
          00116*12
          00117
          00118
          00119*62
          00119*62
          00119*62

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KF=K0F(K)
COEFS=PHIS/(1.-FUIS)
COEFR=1.0
*****
*****
CALL FORMB AND CREATE ELEMENT STIFFNESSES BY GAUSS INTEGRATION
FOR FOUR NODDED ISOPARAMETRIC ELEMENT **MODIFY FOR EIGHT NODDED
*****
CALL FORMTA(KKK, IPLNAX, IKS, IKF, PERM, TTX0, TTX1, TTX2, NUMNP, THICK,
*NEL, WA, PRESS, PS, RS, RF, COEFS, COEFR, FACN)
IF(IJUMP.NE.0) GO TO 615
CALL STDSLD(NP, IH, IH, IH, NUMNP, NEL, FK, DENS, DENF, GMS,
*VOL, K, F, EM, IMAT, KS, KF, THICK)
KS=IKS
KF=IKF
CALL SUBDMA(VOL, EM, IMAT, NS, KF, NEL, THICK)
*****
615 CONTINUE
L=0
DO 620 I=1,4
IIM=I
N=NP(IIM)
DO 620 J=1,4
L=L+1
LM(L)=10*(N,J)
620 CONTINUE
IF(KNS.EQ.2) GO TO 775
WRITE(6,400)
400 FORMAT(5X, ' PRINT IN FLSTIF ')
WRITE(6,401) IMAT, DBETA1(IMAT), DBETA2(IMAT)
401 FORMAT(5X, 15, 2E10.3)

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00120*62
00121*62
00122*63
00123
00124
00125
00126
00127
00128
00129
00130
00131
00132
00133
00134**2
00135*51
00136*62
00137*63
00138*63
00139*63
00140*63
00141**2
00142**2
00143**2
00144*38
00145*33
00146*34
00147**2
00148
00149**3
00150**3
00151**3
00152*26
00153*26
00154**3
00155*12
00156**3
00157**3
00158
00159
00160
00161
00162
00163
00164
00165
00166
00167
00168
00169
00170
00171
00172
00173**3
00174**3
00175*13

```



```

CALL F03DPA(VOL, DENS, DPEF, KS, KE, DDETA1, DDETA2, F, NUMNP, NEL, 00176*17
*ID, IH, IMAT) 00177*12
WRITE(6, 400)
WRITE(6, 401) IMAT, DDETA1(IMAT), DDETA2(IMAT)
*****
00178
00179
00180
00181
00182
***TAPE 13 *** CONTAINS INFORMATION ON ELEMENT CONNECTIVITY AND 00183
STIFFNESS AND MASS / DAMPING MATRICES AND LATER USED IN **KS00184
FOR ASSEMBLY PURPOSES 00185
00186*31
00187*31
***TAPE 11** STRESS-DISPL MATRIX FOR SOLID, STRAIN -DISPL MATRIX FOR 00188
CONNECTIVITY ARRAY, SOLID COMPRESSIBILITY, FLUID COMPRESSIBILITY, AND 00189
LATER USED IN STRESS SUBROUTINE 00190
*****CHANGED FROM ORIGINAL FOR STATIC TRANSIENT PROBLEM**** 00191
00192*31
00193*31
00194*31
***** CALL LOAD ROUTINE TO CALCULATE RESIDUAL STRESSES IN 00195*31
***** TERMS OF INITIAL STRESSES OR NONLINEAR ANALYSIS 00196*31
00197*31
***** PASS ' ** WA ** ' VECTOR AT THIS POINT FOR 00198*32
***** ASPIC ANALYSIS TO CALCULATE THE RESIDUAL LOAD VECTOR 00199*32
00200*32
00201*32
00202*31
00203*50
00204*31
00205*50
00206*50
00207*50
00208*62
00209*13
00210*13
00211*13
00212*40
***** REDEFINE K BECAUSE OF LIN USE IN SUBDID ***** 00213*40
N=IMAT 00214*40
00215*23
WRITE(11) ST, IT, PI, LM, EAL, PA(K), EM(K), RM, ZM, KS, KE, IMAT, BLK, EMU(K), 00216*27
*YOUNG, THIS, VOL, PS, FACN 00217*63
00218
00219*39
00220*41
00221
00222
***** 00223
00224
00225
00226
00227
00228
00229
00230
00231*12
00232*12
***** FIND BANDWIDTH FOR SOLID PART ONLY ***** 00233*12

```

```

      DO 800 I=1,8
      L=LK(I)
      IF(L) 800,800,830
030 DO 850 J=1,8
      JJM=J
      N=LK(JJM)
      IF(N) 850,850,855
055 KL=1AHS(N-1)
      IF(KL.GT.MBANDS) MBANDS=KL
      850 CONTINUE
      800 CONTINUE

CCCC
      FINDS BAND WIDTH
      DO 840 I=1,16
      L=LM(I)
      IF(L) 840,840,830
030 DO 835 J=1,16
      JJM=J
      N=LM(JJM)
      IF(N) 835,835,832
032 KL=1AHS(N-L)
      IF(KL.GT.MBAND) MBAND=KL
035 CONTINUE
040 CONTINUE
      IF(ML) 50,850,40
050 IF(NUMEL-NEL) 50,800,40
      50 WRITE(6,2007) INEL
      STOP

C
060 MBAND=MBAND+1
      MBANDH=MBAND
      MBANDS=MBANDS+1
      IF(BC.GT.0.AND.KKS.EQ.3) GO TO 10
      WRITE(6,2090) MBAND,MBANDH,MBANDS

C
C
C
10 CONTINUE

      IF(KKS.NE.3) GO TO 700
      IF(BC.NE.0) GO TO 70
      DO 80 I=1,NUMMAT
      HPH=(ELMBA(1))/(2.*(ELMBA(1)+ELM(1)))
      PRP(1,1)=EMU(1)*(2.*(1+ENU))
      PRP(2,1)=ENU
      READ(5,1008) (N,(PRP(J,N),J=3,NCOW))

C
      80 CALL PRIP(N,PRP(1,N),NUMMAT)
      70 CONTINUE

CCCCC
700 RETURN

```

```

00234*12
00235*12
00236*14
00237*12
00238*12
00239*13
00240*15
00241*12
00242*12
00243*14
00244*12
00245*27
00246*27
00247*27
00248
00249
00250
00251
00252
00253
00254
00255
00256
00257
00258
00259
00260
00261**3
00262**3
00263**3
00264
00265
00266
00267*12
00268*71
00269*12
00270*48
00271*48
00272*71
00273*48
00274*47
00275*48
00276*44
00277*70
00278*70
00279*70
00280*70
00281*70
00282*44
00283*44
00284*47
00285*55
00286*55
00287*58
00288*58
00289*47
00290
00291
00292
00293

```

C 1008 FORMAT(15,4E12.5)

C 1001 FORMAT(A10)

2090 * BAND WIDTH FOR H = ,110/

* BAND WIDTH FOR S = ,110/

2007 FORMAT(//, ELEMENT CARD ERROR, THEMAXI,15)

END

00294
00295*70
00296
00297
00298*27
00299*12
00300*12
00301**4
00302

```

DATA SET EDTZDO AT LEVEL 020 AS OF 83/02/02
SUBROUTINE EDTZDO(PF,FACN,PS,LT,PMAX,NUMEL)
IMPLICIT REAL*8(A-H,O-Z)

MODEL = 0

ELASTOPLASTIC MODEL (VON MISES)

COMMON/EL/IND,ICOUNT
COMMON/DIMEL/M43
COMMON/MAT/MID/STRESS(9), STRAIN(4),C(4,4),IPT,NEL
COMMON/BELA/NS,NELTYP,ND,NUMGT,MODEL,NCUN,IDW,NK015,NK016,NK017
COMMON/BOBBY/KKS,NUMMAT,NC,LOOP,NDRN,WINE,IDWA,NPT

COMMON D(1)
DIMENSION PMAX(NUMEL,1)
DIMENSION IA(400000)
EQUIVALENCE (D(1),IA(1))
DIMENSION PF(1)
DIMENSION PS(16,8),FACN(4),LM(16)

FOR ADDRESSES NI01,NI02,NI03,... SEE SUBROUTINE TODMFE

WRITE(0,15)
15 FORMAT(5X,1 PRINT IN EDTZDO *)
WRITE(0,20) NS,NELTYP,ND,NUMGT,MODEL,NCUN,IDW,NK015,NK016,NK017
20 FORMAT(5X,10I5)

INITIALIZE WORKING ARRAY
IF(IND.NE.-1) GO TO 100

IDW=10
NN=NK017+(NEL-1)*NPT*IDW
MATP=1A(2*NK015+NEL-2)
NM=NK016+(MATP-1)*4
DO 10 J=1,NPT
10 CALL 1ELPAL(D(NN),D(NN+9+(J-1)*10),D(NM),NPT,J)

WRITE(0,15)
WRITE(0,20) NS,NELTYP,ND,NUMGT,MODEL,NCUN,IDW,NK015,NK016,NK017

RETURN

FIND STRESS-STRAIN LAW AND STRESS

100 CONTINUE
IDW=10

```

```

00001*17
00002
00003
00004
00005
00006
00007
00008
00009**2
00010**2
00011*10
00012**6
00013*13
00014*13
00015**4
00016**4
00017**4
00018*17
00019**4
00020**4
00021*14
00022*15
00023**5
00024
00025
00026
00027
00028
00029*19

00030*19
00031*19
00032
00033
00034*16
00035*16
00036**5
00037**8
00038**5
00039**5
00040**5
00041**5
00042*12

00043*12
00044
00045**9
00046
00047
00048
00049
00050
00051
00052**4
00053**8

```



```

DATA SET FLT207      AT LEVEL 020 AS OF 83/01/24
SUBROUTINE ELT2D7(PF, FACN, PS, LM, PMAX, NUMEL)
IMPLICIT REAL*8(A-H, O-Z)

MODEL = 7

ELASTOPLASTIC MODEL (DRUCKER-PRAGER)

COMMON/EL/IND, ICOUNT
COMMON/DIMEL/N43
COMMON/BELA/NG, NELTYP, ND, NUMGT, MODEL, NCON, IDW, NK015, NK016, NK017
COMMON/MATMOD/STRESS(9), STRAIN(4), C(4,4), IPT, NEL
COMMON/BUBBY/KKS, NUMMAT, NC, LOOP, NDRN, NINT, IDWA, NPT

COMMON I(1)
DIMENSION IA(400000)
EQUIVALENCE (O(1), IA(1))
DIMENSION PMAX(NUMEL, 1)
DIMENSION PF(1)
DIMENSION PS(16, 8), FACN(4), LM(16)

FOR ADDRESSES N101, N102, N103, ... SEE SUBROUTINE TDDMFE

MATP=IA(N107 + NEL - 1)

MATP=IA(2*NK015+NEL-2)
NM=NK016+(MATP-1)*NCON

IF(IND.NE.-1) GO TO 100

INITIALIZE WORKING ARRAY
NN=NK017+(NEL-1)*NPT+IDW
NG=NN-1

DO 10 J=1, NPT

```

```

00001*25
00002
00003
00004
00005
00006
00007
00008
00009**6
00010**6
00011*18
00012**6
00013*20
00014*20
00015*26
00016**7
00017**7
00018*18
00019*18
00020*18
00021**2
00022**2
00023*25
00024**7
00025*23
00026
00027
00028
00029
00030
00031*16
00032*16
00033*16
00034*16
00035*16
00036*18
00037*16
00038*18
00039*18
00040*11
00041*11
00042*24
00043
00044
00045
00046
00047**9
00048*13
00049*21
00050*13
00051*13
00052*13
00053*13
00054*13
00055
00056*10
00057*11
00058*11
00059*21

```

CALL IDRUCK(D(NH),D(NN+10+(J-1)*11),D(NM),NPI,J)	00060*22
	00061*10
10 CONTINUE	00062*10
	00063*10
RETURN	00064
	00065
	00066
	00067
FIND STRESS - STRAIN LAW AND STRESS	00068
	00069
	00070
100 NS=NK017+((NLL-1)*NPI+(IPT-1))*IDW	00071**6
	00072
	00073*18
	00074*21
	00075*22
	00076*22
	00077*22
	00078*18
CALL DRUCK(D(NM),D(NS),D(NS+4),D(NS+8),D(NS+9),D(NS+10),PF,PS,FACN	00079*23
*,LM,PMAX,NUMEL)	00080*25
	00081*25
RETURN	00082
	00083
	00084
	00085
END	00086

```

DATA SET EQCHK      AT LEVEL 004 AS OF 83/03/05
SUBROUTINE EQCHK(XM,AS,B,NEQ,MBAND,X0,X1,X2,R,NC,A3,ND00001
* YN,GB,AU,AS)
                                00002**6
                                00003
                                00004
                                00005
      IMPLICIT REAL * 8(A-H,O-Z)
COMMON/BOBBY/KKS
DIMENSION XM(1),AS(NEQ,1),B(NEQ),X0(1),X1(1),X2(1),R(NEQ)
*,A3(NEQ,1),AU(1)
                                00006
                                00007**2
                                00008
                                00009**6
DIMENSION GB(1)
DIMENSION AS(NEQ,1)
***** READ STIFFNESS MATRIX FROM TAPE *****
                                00010
REWIND 60
                                00011
READ(60) ((A3(1,J),J=1,MBAND),I=1,NEQ)
                                00012
                                00013
                                00014
56  WRITE(6,56)
    FORMAT(1H1/5X,' PRINT IN EQCHK ')
    WRITE(6,58)((A3(1,J),J=1,MBAND),I=1,NEQ)
    PRINT 999
    WRITE(6,50)((A5(1,U),U=1,MBAND),I=1,NEQ)
    PRINT 999
    WRITE(6,57)(XM(I),I=1,NEQ)
    PRINT 999
    WRITE(6,57)(X0(I),I=1,NEQ)
    PRINT 999
    WRITE(6,57)(X1(I),I=1,NEQ)
    PRINT 999
999  FORMAT(//)
    WRITE(6,57)(X2(I),I=1,NEQ)
57  FORMAT(5X,12E10.3)
58  FORMAT(5X,8E10.3)
                                00015
                                00016**3
                                00017
                                00018
                                00019
    DO 50 I=1,NEQ
    B(I)=0.0
    R(I)=0.0
50
    DO 51 I=1,NEQ
    DO 51 J=1,NEQ
51  AS(I,J)=0.0
                                00020
                                00021
70  WRITE(6,70)
    FORMAT(1H1/5X,' OVERALL EQUILIBRIUM CHECK 1/ )
                                00022**8
                                00023**8
                                00024
                                00025
                                00026
                                00027
                                00028
                                00029
                                00030
                                00031
                                00032
                                00033
                                00034
    IF(NDYN.EQ.4) GO TO 60
    COMPUTE INERTIA TERM
    DO 1 I=1,NEQ
    B(I)=B(I)+XM(I)+X2(I)
1
    WRITE(6,2)
    FORMAT(5X,' PRINT OF INERTIA TERM ')
    WRITE(6,3) (B(I),I=1,NEQ)
2
3  FORMAT(5X,12E10.3)

```


		00035
	TRANSFER ARRAY	00036
		00037
5	DO 5 I=1,NEQ	00038
	R(I)=B(I)	00039
		00040
55	DO 55 I=1,NEQ	00041
	B(I)=0.0	00042
		00043
		00044
		00045
	COMPUTATION DISSIPATION FORCE	00046
		00047
4	DO 4 I=1,NEQ	00048
	B(I)=B(I)+A5(I,1)*X1(I)	00049
		00050
		00051
		00052
	DO 6 I=1,NEQ	00053
	LN=I-1	00054
	NR=MINO(MBAND,NEQ-1+1)	00055
	IF(NR.EQ.1) GO TO 7	00056
	DO 8 J=2,NR	00057
	M=LN+J	00058
	B(I)=B(I)+A5(I,J)*X1(M)	00059
8	B(M)=B(M)+A5(I,J)*X1(I)	00060
	CONTINUE	00061
6	CONTINUE	00062
7	CONTINUE	00063
		00064
	WRITE(6,9)	00065
9	FORMAT(5X,' PRINT OF DAMPING TERM')	00066
	WRITE(6,3)(B(I),I=1,NEQ)	00067**4
		00068
		00069
10	DO 10 I=1,NEQ	00070
	R(I)=R(I)+B(I)	00071
		00072
60	CONTINUE	00073
	COMPUTATION OF STIFFNESS TERM	00074
		00075
80	DO 80 I=1,NEQ	00076
	B(I)=0.0	00077
		00078
		00079
12	DO 12 I=1,NEQ	00080
	B(I)=B(I)+A3(I,1)*XU(I)	00081
		00082
		00083
		00084
	DO 13 I=1,NEQ	00085
	LN=I-1	00086
	NR=MINO(MBAND,NEQ-1+1)	00087
	IF(NR.EQ.1) GO TO 14	00088
	DO 15 J=2,NR	00089
	M=LN+J	00090
	B(I)=B(I)+A3(I,J)*XU(M)	00091
15	B(M)=B(M)+A3(I,J)*XU(I)	00092
	CONTINUE	00093
13	CONTINUE	00094
14	CONTINUE	

```

10      WRITE(6,10)
      FORMAT(5X,' PRINT OF STIFFNESS TERM  ')
      WRITE(6,3)(B(I),I=1,NEQ)
      ***** TOTAL RESISTING FORCE AT EACH PRINT INTERVAL *****
      DO 100 I=1,NEQ
100     B(I)=B(I)+R(I)
      WRITE(6,200)
200     FORMAT(5X,' TOTAL RESISTIVE FORCE  ')
      WRITE(6,28) NC
      WRITE(6,3) (B(I),I=1,NEQ)

28      FORMAT(5X,' TIME STEP INTERVAL  = ',15/)
      ***** ALSO CHECK DEFINITENESS OF MATRIX FOR COUPLING *****
      IF(KKS.EQ.3) GO TO 310

      DEF=0.0
      DO 300 I=1,NEQ
300     DEF=DEF+XU(I)*B(I)
      CONTINUE
      WURK=0.0
      DO 700 I=1,NEQ
700     WURK=WURK+GB(I)*XU(I)

      DO 350 I=1,NEQ
350     R(I)=1.0
          R(I)=0.0
      CONTINUE

      DO 355 I=1,NEQ
355     R(I)=R(I)+A3(I,1)*AU(1)
      CONTINUE
      DO 360 I=1,NEQ
360     LQ=I-1
          MR=MINO(NBAND,NEQ-1+I)
          IF(MR.EQ.1) GO TO 365
          DO 370 J=2,MR
370     M=LQ+J
          R(I)=R(I)+A3(I,J)*AU(M)
          R(M)=R(M)+A3(1,J)*AU(1)
      CONTINUE
360     CONTINUE
365     CONTINUE
      SUM=0.0
      DO 375 I=1,NEQ
375     SUM=SUM+R(I)*AU(1)
      CONTINUE

      IF(KKS.NE.3) GO TO 119

```

```

00095
00096
00097
00098**4
00099
00100
00101
00102
00103
00104
00105
00106
00107
00108
00109
00110
00111
00112
00113
00114
00115
00116**5
00117**5
00118**5
00119**5
00120**5
00121**6
00122**6
00123**6
00124**6
00125**7
00126**9
00127**9
00128**9
00129**9
00130**9
00131**9
00132**9
00133**9
00134**9
00135**9
00136**9
00137**9
00138**9
00139**9
00140**9
00141**9
00142**9
00143**9
00144**9
00145**9
00146**9
00147**9
00148**9
00149**9
00150**9

```

```

IF(SUM.LE.0.0) GO TO 115
CALL STIFF(A3,NEG,MBAND,A5)

JURN=10
CALL EIGR5(A5,NEG,JURN,R,B,NEG,A0,IER)
WRITE(6,110) (R(I),I=1,NEG)
WRITE(6,111) IER
110  FORMAT(12E10.3)
111  FORMAT(5X,15)
     WORK=R(1)

119  CONTINUE
     IF(WORK.LE.0.0) GO TO 400

115  RATIO=DEF/WORK
     CONTINUE
     IF(SUM.GT.0.0) GO TO 400
     WRITE(6,500) NC,SUM

400  STOP
     CONTINUE

600  WRITE(6,600) NC,DEF,WORK,RATIO,SUM
     FORMAT(5X,' ***** TIME STEP (NC) = ',15/
*      5X,' ***** STIFFNESS (DEF) = ',E10.3/
*      5X,' ***** WORK TERM (WORK) = ',E10.3/
*      5X,' RATIO = ',E10.3/
*      5X,' DEFINITENESS = ',E10.3/)
500  FORMAT(5X,' TIME STEP NC = ',15/
*      5X,' VALUE OF MATRIX A3 = ',E10.3/)
310  CONTINUE
     RETURN
     END

```

00151**9
00152**7

00153**6

00154**9
00155**9
00156**9

00157**5

00158**5

00159**7

00160**9

00161**7

00162**7

00163**7

00164**9

00165**9

00166**5

00167**5

00168

00169

```

DATA SET ESPDMA AT LEVEL 025 AS OF 83/03/05
SUBROUTINE ESPDMA(VOL,DENS,DENF,KS,KF,DBT1,DBT2,F,NUMNP,NEL,LD,IH,
*IMAT)
IMPLICIT REAL*8 (A-H,U-Z)
DIMENSION ID(NUMNP,1),X(1),Y(1),ELMDA(1),EMD(1),EALFA(1),EM(1),
*EK(1),RUS(1),RUF(1),NP(4),INP(4),TTI(2),SSS(2),IS(8),IL(8),IH(NUMN
*P,1)
COMMON/CONT/DT,HED(8),RADN,NASIM,NOMEL,NEG,MBAND,NCYCL,NDYN,IPLNA
*X,ISTRPK,NPRINT,ISTARI,NEGH,MBANDH,NEGS,MBANDS
COMMON/BOBBY/KKS,NUMMAT,NC,LOOP,NDRN,NINT,IDWA,NPT
COMMON/EMB/S(16,16),GM(16),GH(16,16),SS(16,16),XP(16),XN(
*16),LN(16)
COMMON/QUAD/RK(4),ZZ(4),FAC,R,U(6),P(4,12),SF(12,12),H(12,12),
*ST(20,12),D(4,4),C(12,12),E(12,12),TT(5,12),PI(4,8),ODD(4,4)
COMMON/ASHK/C1,C2,C3,AL,ALF,FACC,FACT
DIMENSION LK(8),LH(8),FK(1),DBT1(1),DBT2(1)
DATA IS/1,2,5,6,9,10,13,14/,IL/3,4,7,8,11,12,15,16/
FORM ELEMENT LOCAL STIFFNESS MASS DAMPING MATRICES TAKING *1
DEGREES OF FREEDOM
K=IMAT
WRITE(6,60)K,IMAT,LOOP,DBT1(K),DBT2(K)
60 FORMAT(5X,3I5,2E10.3)
VLD=VOL/4.
DO 626 I=1,8
NS=IS(I)
NF=IL(I)
IF(KS.EQ.-1.AND.KF.EQ.1) GO TO 656
GM(NS)=VLD*DENS
IF(KS.EQ.1.AND.KF.EQ.-1) GO TO 658
656 GM(NF)=VLD*DENF
658 CONTINUE
DO 625 J=1,8
MS=IS(J)
MF=IL(J)
*****
IF(KS.EQ.1.AND.KF.EQ.-1) GO TO 616
IF(KS.EQ.0.AND.KF.EQ.0) GO TO 616
IF(KS.EQ.-1.AND.KF.EQ.1) GO TO 619
*****

```

```

00001*10
00002*13
00003*10
00004*13
00005
00006*13
00007
00008
00009
00010*13
00011*12
00012
00013
00014
00015
00016*11
00017*23
00018*23
00019*20
00020*20
00021*17
00022*17
00023
00024
00025
00026**8
00027
00028
00029
00030*17
00031
00032
00033
00034
00035**8
00036*14
00037*25
00038*14
00039
00040
00041
00042
00043*25
00044
00045*25
00046
00047*14
00048*14
00049
00050
00051
00052*14
00053
00054
00055
00056
00057

```

```

616 S(NS,MS)=SF(1,J)
      IF(LOOP.NE.1) GO TO 660
      GH(NS,MS)=DBT1(K)*(GM(MS)-(F**2.)*GM(MF))+DBT2(K)*(SF(1,J)-E(I,J)
**ALF)
      GO TO 665
660 CONTINUE
      GH(NS,MS)=DBT1(NEL)*(GM(MS)-(F**2.)*GM(MF))+DBT2(NEL)*(SF(I,
**J)-E(1,J)*ALF)
665 CONTINUE
      IF(KS.EQ.1.AND.KF.EQ.-1) GO TO 625
618 S(NS,MF)=C(1,J)
      S(NF,MS)=C(J,1)
619 GH(NF,MF)=H(1,J)
617 S(NF,MF)=E(1,J)
      SS(NS,MS)=SF(1,J)
625 CONTINUE
626 CONTINUE
*****
999 FORMAT(//)
      IF(NC.GT.0.AND.KKS.EQ.3) GO TO 8200
      IF(NEL.NE.0) GO TO 8200
      PRINT 999
      WRITE(6,9511)
9511 FORMAT(1H1/' ***** CONNECTIVITY ARRAY ***** '/')
      WRITE(6,220) (LM(1),1=1,16)
220 FORMAT(10X,16I5)
      PRINT 999
      WRITE(6,9513)
9513 FORMAT(5X,' ELEMANT UNDRAINED STIFFNESS MATRIX')
      WRITE(6,221) ((S(1,J),J=1,16),1=1,16)
      PRINT 999
      WRITE(6,9514)
9514 FORMAT(5X,' ELEMENT MASS MATRIX')
      WRITE(6,224) (GM(1),1=1,16)
224 FORMAT(3 X,16E8.1)

```

```

00058*14
00059
00060*24
00061*24
00062*24
00063*24
00064*24
00065*24
00066*24
00067*24
00068*24
00069*24
00070*24
00071*24
00072*24
00073*24
00074*24
00075*24
00076*24
00077*24
00078
00079*14
00080*14
00081
00082
00083
00084
00085*19
00086
00087*14
00088*22
00089*22
00090*22
00091
00092
00093
00094
00095
00096
00097
00098
00099
00100*18
00101*18
00102*18
00103
00104
00105
00106*16
00107
00108
00109
00110
00111
00112
00113
00114
00115
00116
00117

```

	PRINT 999	00118
	WRITE(6,9515)	00119
9515	FORMAT(5X,'ELEMENT DISSIPATION MATRIX')	00120
	WRITE(6,221)((GH(I,J),J=1,10),I=1,10)	00121
221	FORMAT(3X,10F8.1)	00122
	PRINT 999	00123
	WRITE(6,9517)	00124
9517	FORMAT(5X,'ELEMENT UNDRAINED STIFFNESS MATRIX ASSUMING FLUID *HAS NO DISPLACEMENT')	00125
	WRITE(6,221)((SS(I,J),J=1,10),I=1,10)	00126
	PRINT 999	00127
	WRITE(6,9518)	00128*21
9518	FORMAT(5X,'ELEMENT INITIAL STRESS LOAD FACTOR')	00129*21
	WRITE(6,221)(XP(I),I=1,10)	00130*21
	PRINT 999	00131*21
	WRITE(6,9519)	00132*21
9519	FORMAT(5X,'ELEMENT SELF WT. LOAD VECTOR')	00133*21
	WRITE(6,221)(XN(I),I=1,10)	00134*21
8200	CONTINUE	00135*21
	RETURN	00136
	END	00137
		00138

```

DATA SET ESFMTA AT LEVEL 042 AS OF 83/02/11
SUBROUTINE ESFMTX(XKR, IPLNAX, IKS, IKF, PERM, TTX0, TTX1, TTX2, NUMNP,
* THICK, NED, WA, PRESS, PS, RS, RT, COEFS, COEPR, FACN)
IMPLICIT REAL*8(A-H, O-Z)
COMMON/QUAD/HR(*), ZZ(*), FAC, N, NTO, I(*, 12), ST(12, 12), H(12, 12),
*SI(20, 12), D(4, 4), C(12, 12), E(12, 12), TT(5, 12), PI(4, 8)
COMMON/BELTA/NS, NELIYP, ND, NUMGT, MODEL, NCON, IDW, NK015, NK016, NK017
COMMON/ASHK/C1, C2, C3, AD, AH, FAC, FACT
COMMON/BOBBY/KKS, NUMMAT, NC, LOOP, NDRN, NINT, IDWA, NPT
COMMON/RANI/RM(5), ZM(5), IPT
COMMON/INTEGE/ETA, BEIA, GAMA, KTOI, NGAMA, NPUNCH, NEQUIB, ITEMX
COMMON/GORA/IFLAG, ITR
DIMENSION X(1), Y(1), ELMDA(1), EMU(1), EALFA(1), EM(1),
*EK(1), ROS(1), ROF(1), NP(4), INP(4), TTT(2), SSS(2), IS(8), IL(8)
DIMENSION TT1(12), TT2(12), TTX1(2, 12), TTX0(12, 2), TTX2(12, 12)
DIMENSION LK(8), LH(8)
DIMENSION PRESS(5)
DIMENSION WA(IDWA, 1)
DIMENSION PS(16, 8)
DIMENSION FACN(4)
DATA BSS/-1.000, 1.000/, TTT/-1.000, 1.000/
*****
*****
CALL FORMB AND CREATE ELEMENT STIFFNESSES BY GAUSS INTEGRATION
FOR FOUR NODDED ISOPARAMETRIC ELEMENT **MODIFY FOR EIGHT NODDED
DU 510 J=1,4
DU 510 I=1,8
510 PI(I, J)=0.0
DU 520 I=1,16
DU 520 J=1,8
520 PS(I, J)=0.0

```

```

00001**4
00002*13
00003*32
00004*13
00005
00006*13
00007*10
00008*10
00009*42
00010*42
00011
00012*22
00013*24
00014*40
00015*40
00016*33
00017*35
00018*35
00019*31
00020*31
00021*42
00022*42
00023*31
00024**5
00025*20
00026*20
00027*20
00028**5
00029*13
00030
00031
00032*10
00033**4
00034
00035
00036*20
00037*26
00038*20
00039*28
00040*32
00041*28
00042*28
00043*12
00044
00045
00046
00047
00048
00049
00050
00051*18
00052*18
00053**6
00054**6
00055**6
00056
00057*36
00058*36
00059*36

```

```

KS=IKS
KF=IKF
DO 500 LR=1,NINT
SX=SSS(LR)*XKK
DU 500 LZ=1,NINT
TX=TTT(LZ)*XKK
CALL FURMB(SX,TX,IPLNAX,THICK)

IPT=(LR-1)*NINT+LZ
RM(IPT)=RM(IPT)+Q(1)*RR(1)+Q(2)*RR(2)+Q(3)*RR(3)+Q(4)*RR(4)
ZM(IPT)=ZM(IPT)+Q(1)*ZZ(1)+Q(2)*ZZ(2)+Q(3)*ZZ(3)+Q(4)*ZZ(4)

IF(NC.EQ.0) GO TO 100
CALL MODCNV(NEL,IPT,D)

100 CONTINUE
WRITE(6,101)
101 FORMAT(1H1/5X,' PRINT IN EBFMTX.FOR ')
WRITE(6,102) ((b(i,j),j=1,4),i=1,4)
102 FORMAT(5X,4E10.3)

*****
IF(KS.EQ.0.AND.KF.EQ.0) GO TO 80
IF(KS.EQ.1.AND.KF.EQ.-1) GO TO 86
IF(KS.EQ.-1.AND.KF.EQ.1) GO TO 95
*****

***** REPLACE C1,C2,C3, *** BY D(I,J) FOR PLANE S00093*13
***** F L A S T O P L A S T I C A N A L Y S I S WE NEED 00094*14
***** MATRICES ANYWAY ***** 00095*14

80 DO 90 I=1,ND
***** CHANGED FOR GENERAL PURPOSES *****00101*15
*****
D1=(D(1,1)*P(1,1)+D(1,2)*P(2,1)+D(1,3)*P(3,1)+D(1,4)*P(4,1))*FAC 00104*15
D2=(D(2,1)*P(1,1)+D(2,2)*P(2,1)+D(2,3)*P(3,1)+D(2,4)*P(4,1))*FAC 00105*15
D3=(D(3,1)*P(1,1)+D(3,2)*P(2,1)+D(3,3)*P(3,1)+D(3,4)*P(4,1))*FAC 00106*15
D4=(D(4,1)*P(1,1)+D(4,2)*P(2,1)+D(4,3)*P(3,1)+D(4,4)*P(4,1))*FAC 00107*15
*****
DO 90 J=1,ND
SF(J,1)=SF(J,1)+D1*P(1,J)+D2*P(2,J)+D3*P(3,J)+D4*P(4,J) 00112
SF(1,J)=SF(J,1) 00113
***** 00114
***** 00115*13

```



```

                                00116*13
                                00117*13
IF(KS.EQ.1.AND.NF.EQ.-1) GO TO 5000
                                00118**6
95 FACK=FAC/PERM
                                00119
                                00120**9
                                00121**9
                                00122*11
***** PARTION REMOVED FROM ORIGINAL VERSION *****
                                00123*11
                                00124*11
                                00125*11
                                00126*11
                                00127**9
                                00128**9
                                00129*13
                                00130*13
IF(NELTYP.EQ.0) NN=4
                                00131*13
IF(NELTYP.EQ.1) NN=6
                                00132*13
                                00133*13
                                00134*13
IF(NDRM.EQ.0) GO TO 5000
                                00135*17
**** FURN FLUID DISSIPATION MATRIX *****
                                00136*13
                                00137*13
                                00138*13
I1=1
                                00139
JJ=2
                                00140
L=0
                                00141
                                00142*14
***** SHAPE FUNCTION *****
                                00143*14
DU 2100 J=1,NN
                                00144*14
DU 2101 J=1,2
                                00145
L=L+1
                                00146
TT1(L)=0.0
                                00147
TT2(L)=0.0
                                00148
IF(L.EQ.I1) TT1(L)=Q(1)
                                00149
IF(L.EQ.JJ) TT2(L)=Q(1)
                                00150
2101 CONTINUE
                                00151
I1=I1+2
                                00152
JJ=JJ+2
                                00153
2100 CONTINUE
                                00154
                                00155*14
                                00156*14
DU 2010 J=1,ND
                                00157*10
2010 TTX1(1,J)=TT1(J)
                                00158
DU 2020 J=1,ND
                                00159*10
2020 TTX1(2,J)=TT2(J)
                                00160
                                00161*14
                                00162*14
***** PERMIABILITY MATRIX FORMATION *****
                                00163*14
                                00164*14
DU 205 I=1,2
                                00165
DU 205 J=1,ND
                                00166*10
205 TTX0(J,I)=TTX1(I,J)
                                00167
DU 206 I=1,ND
                                00168*10
DU 206 J=1,ND
                                00169*10
DU 207 K=1,2
                                00170
207 TTX2(I,J)=TTX2(I,J)+TTX0(I,K)*TTX1(K,J)
                                00171
H(I,J)=H(I,J)+TTX2(I,J)*FACK
                                00172
206 CONTINUE
                                00173
                                00174**9
                                00175**9

```

```

5000 CONTINUE                                00176**9
*****00177
00178*20
00179*20
***** CALCULATION OF RESIDUAL LOAD VECTOR AT GAUSS POINT 00180*20
**** ELASTOPLASTIC ANALYSIS 00181*20
00182*20
00183*39
CALL LOAD( NEL,PRESS,VOL,WA) 00184*20
00185*20
00186*20
00187*29
00188*29
00189*29
DO 700 I=1,ND 00190*20
DO 700 JJ=1,4 00191*20
J=JJ+(I-1)*4 00192*20
PS(J,I)=P(JJ,I) 00193*20
700 CONTINUE 00194*32
FACN(I)=P(JJ,I) 00195*32
00196*20
00197*20
00198*20
600 CONTINUE 00199*26
00200*30
00201*30
00202*31
00203*31
00204*31
IF(NGAMA.EQ.0) GO TO 500 00205*30
***** CALL SELFWT SUBROUTINE ***** 00206*32
00207*32
00208*32
550 CONTINUE 00209*30
00210*41
IF(IFLAG.NE.0) GO TO 500 00211*41
00212*41
CALL SELFWD (NEL,RS,RF,COEFS,COEFR,VOL) 00213*30
00214*30
00215*30
00216**8
00217
500 CONTINUE
WRITE(6,701)
701 FORMAT('PRINT INDEPMIX STRAIN MATRIX VOLUME AT GAUSS POINTS ')
WRITE(6,702) ((PS(I,J),J=1,8),I=1,16)
WRITE(6,702) (FACN(I),L=1,4)
702 FORMAT(5X,8E10.3)
00218*16
00219*16
00220*16
RETURN 00221
END 00222

```

```

DATA SET IDRUCK      AT LEVEL 005 AS OF 82/03/31
SUBROUTINE IDRUCK(WA,IWA,PROP,NPT,J)
IMPLICIT REAL*8(A-H,O-Z)
COMMON/BELA/NS,NELTYP,ND,NUMGT,MODEL,NCON,IDW,NK015,NK016,NK017
DIMENSION WA(IDW,1),PROP(NCON)

SET INITIAL STRESSES AND STRAINS EQUAL TO ZERO
DO 15 I=1,8
WA(I,J)=0.0
15 CONTINUE

ANGLE=PROP(3)  AND  CINHES=PROP(4)

WA( 9,J)=PROP(3)/57.296
WA(10,J)=PROP(4)

SET INITIAL STRESS STATE TO -ELASTIC-
IWA=1

25 CONTINUE

RETURN

END

```

```

00001**4
00002
00003**4
00004
00005**4
00006
00007
00008
00009
00010
00011
00012
00013
00014
00015
00016
00017
00018
00019
00020
00021
00022**4
00023
00024
00025
00026
00027
00028
00029

```

	DATA SET IELPAL	AT LEVEL 006 AS OF 82/03/31	00001	
	SUBROUTINE IFLPAL(WA, IWA, PROP, NPT, J)		00002	
	IMPLICIT REAL*8(A-H, O-Z)		00003	
	COMMON/BELA/NS, NELTYP, ND, NUMGT, MODEL, NCON, IDW, NK015, NK016, NK017		00004	00004**2
	DIMENSION WA(IDW, 1), PROP(NCON)		00005**2	
	SET INITIAL STRESSES AND STRAINS TO ZERO		00006	
	SET INITIAL YIELD POINT TO PROP(3)		00007	
	SET INITIAL STRESS STATE TO *ELASTIC*		00008	
			00009	
	WRITE(6, 160)			
160	FORMAT(5X, ' *** PRINT IN IELPAL *** ')			
161	WRITE(6, 161) NS, NELTYP, ND, NUMGT, MODEL, NCON, IDW, NK015, NK016, NK017			
	FORMAT(5X, 10I5)			
			00010	
			00011	
	DO 15 I=1, 8		00012**2	
15	WA(1, J)=0.0		00013**2	
	WA(9, J)=PROP(3)		00014	
	IWA=1		00015	
			00016	
25	CONTINUE		00017**2	
			00018**5	
	RETURN		00019	
	END		00020	

```

DATA SET INFMTA      AT LEVEL 056 AS OF 82/07/08
SUBROUTINE INFMTX(S,C,CT,AN,ANIT,AS,SI,SM,SN,SP,SD,SQ,NPJ,NPK,
*NEQS,A,CE,PSOL,DSOL,RW,IW,AM,EK,IK,IH,CUPI,ICOLMS,IDES,ROW,
*GB,SU,PPSUL,DDSUL,SB,XP)
IMPLICIT REAL*8(A-H,O-Z)

DIMENSION S(NPJ,1),C(NPJ,1),CT(NPJ,1),SI(NEQS),AS(NEQS,1),SM(NEQS)
*,SN(1),SP(1),SU(1),SD(NPJ,1)
DIMENSION AN(NPJ,1),ANIT(NPJ,1)

COMMON/ASIM/ NYP,NSC,NSLC,NLC,NNYP,NNSC
COMMON/CONTL/DD,HED(8),RADN,NUMNP,NUMEL
COMMON/TARA/LPC,N,M1,M2,NPT,KPT,NPRNT,M,MM
COMMON/BUDHU/NDIV,NNUD,JK(20,3)

DIMENSION A(LPC,1),PSOL(1),DSOL(1),RW(LPC),IW(1),AM(1),EK(1)
DIMENSION CE(1)
DIMENSION PPSUL(1),DDSUL(1),SB(1),IH(NUMNP,1)
DIMENSION GB(1),SU(1),CUPI(NPK,NPK),ICOLMS(1),IDES(1),ROW(1)

DIMENSION XP(NPK,1)
999 FORMAT(//)

***** INITIALIZE
DO 10 K=1,NEQS
SM(K)=0.0
10 SI(K)=0.0
DO 20 J=1,NPJ
SN(J)=0.0
SP(J)=0.0
20 CONTINUE

MN=(NPK+1)+2
IF(NPT.EQ.1) MN=NPK+NEQS+1+2
IF(NPT.EQ.2) MN=NPK+NEQS+1+2

IF(KPT.EQ.1) MN=MM
DO 40 I=1,MN
40 SU(I)=0.0

**** INITIALIZE ***
DO 650 I=1,LPC
DO 650 J=1,N
650 A(I,J)=0.0
650 CONTINUE

```

```

00001**4
00002*28
00003*54
00004*55
00005*54
00006*28
00007
00008*28
00009*28
00010*43
00011*11
00012*26
00013*26
00014
00015*13
00016*13
00017*28
00018*28
00019*32
00020*56
00021*13
00022*29
00023*13
00024*13
00025*17
00026*30
00027*54
00028*13
00029
00030*54
00031*54
00032**8
00033*11
00034**5
00035**5
00036*11
00037**5
00038**5
00039**5
00040**5
00041**5
00047*19
00048*19
00049*32
00050*45
00051*45

00052*19
00053*19
00054*11
00055*13
00056*13
00057*13
00058*13
00059*13
00060*11
00061*13
00063*32

```

```

***** BRANCH HERE FOR STRESS OPTIMIZATION W. R. T. RESI
* STRESSES *****
IF(KPT.EQ.2.AND.NP1.EQ.1.OR.KPT.EQ.2.AND.NPT.EQ.2) GO TO 480
IF(KPT.EQ.1.AND.NP1.EQ.2.OR.KPT.EQ.1.AND.NPT.EQ.1) GO TO 495

DU 50 J=1,NPJ
DU 30 K=1,NPK
SD(J,K)=0.0
30 CONTINUE
50 CONTINUE

***** COMPUTATION OF 'SC' MATRIX *****
PRINT 999
DU 100 I=1,NPJ
DU 200 K=1,NEQS
SUM=0.0
DU 300 J=1,NPJ
300 SUM=SUM+S(I,J)*C(J,K)
200 SI(K)=SUM

**** COMPUTE 'S * C * (K) * -1' MATRIX (FIRST ROW) *****
DU 400 K=1,NEQS

SUM=0.0
DU 500 L=1,NEQS
500 SUM=SUM+SI(L)*AS(L,K)
SM(K)=SUM
400 CONTINUE

**** COMPUTE 'S * C * K * -1 CT' MATRIX (FIRST ROW) *****
DU 600 J=1,NPJ
SUM=0.0
DU 700 K=1,NEQS
700 SUM=SUM+SM(K)*CT(J,K)
600 SN(J)=SUM

**** COMPUTE 'C * S * K * -1 * CT * S' MATRIX *****
DU 800 J=1,NPJ
SUM=0.0
DU 900 L=1,NPJ
900 SUM=SUM+SN(L)*S(L,J)
800 SP(J)=SUM

***** COMPUTE 'S * C * S * -1 * CT * S - >' MATRIX (Z- MATRIX)
DU 1000 J=1,NPJ
1000 SP(J)=SP(J)-S(I,J)

```

```

00064*32
00065*32
00066*33
00067*32
00042*27
00043*25
00044*27
00045*12
00046*27
00070*44
00071*44
00072*23
00073*28
00074
00075
00076
00077
00078*11
00079
00080
00081
00082
00083
00084
00085
00086*11
00087*11
00088*11
00089*11
00090*17
00091*17
00092*17
00093*11
00094*11
00095*11
00096*11
00097*11
00098*11
00099*11
00100*11
00101*11
00102*11
00103*43
00104*11
00105*11
00106*11
00107*11
00108*11
00109*11
00110*11
00111*19
00112*11
00113*11
00114*11
00115*11
00116*11

```

```

DO 1010 J=1,NPJ
ANIT(1,J)=SP(J)
1010 CONTINUE

```

```

***** TEST FOR SEMIDEFINITENESS OF MATRIX Z *****

```

```

SUM=0.0
DO 150 J=1,NPJ
150 SUM=SUM+SP(J)
SD(1)=SUM

```

```

***** COMPUTATION OF ' Z * N ' MATRIX *****

```

```

PRINT 999
DO 1700 K=1,NPK
SUM=0.0
DO 1700 J=1,NPJ
1700 SUM=SUM+SP(J)*AN(J,K)
1900 SD(1,K)=SUM
100 CONTINUE

```

```

00117*11
00118*13
00119*13
00120*13
00121*13
00122*14
00123*13
00124*13
00125*13
00126*13
00127
00128
00129
00130**3
00131**8
00132
00133
00134
00135*26
00136*26
00137*11
00138*13

```

```

254 WRITE(6,254)
FORMAT(1H1/5X,' PRINT OF Z MATRIX '/')
WRITE(6,251) ((ANIT(I,J),J=1,NPJ),I=1,NPJ)
281 FORMAT(5X,12E10.3)

```

```

JOBN=10

```

```

CALL EIGRS(ANIT,NPJ,JOBN,RW,XP,NPK,SO,IER)
WRITE(6,255) (RW(1),I=1,NPJ)
WRITE(6,256) IER
256 FORMAT(5X,15)

```

```

SUM=0.0
DO 160 J=1,NPJ
160 SUM=SUM+SD(J)
IF(SUM.LT.0.0) GO TO 180
WRITE(6,190)
STOP
15 CONTINUE

```

```

00139*13
00140*13
00141*13
00142*13
00143*13
00144*13
00145*13
00146*41
00147*13
00148*13

```

```

***** INPUT VALUE OF HARD *****

```

```

***** COMPUTATION OF ' N ** T * Z * N ' MATRIX *****

```

```

180 CONTINUE

```

```

00150
00151**3
00152*13
00153*13

```

HARD=0.0

~~DO 4000 I=1,NPK~~
~~DO 5000 J=1,NPK~~
 SUM=0.0
 SUM1=0.0

00154*11
 00155*11
 00156*11
 00157*11

1200 DO 1200 K=1,NPJ
 SUM1=SUM1+AN(K,1)*AN(K,J)
 SUM1=HARD*SUM1

~~DO 1100 K=1,NPJ~~
~~SUM=SUM+AN(K,1)*SD(K,J)~~
 IF(KPT.EQ.1) GO TO 60

00158*11
 00159*26

A(1,J+1)= SUM1-SUM
 GO TO 5000

00160*22

60 CONTINUE
 A(1+2,J)=(SUM1-SUM)

5000 CONTINUE
 4000 CONTINUE

00161*11
 00162*11
 00163
 00164*13
 00165*13

~~DO 220 I=1,NPK~~
 IF(KPT.EQ.1) GO TO 222

00167*13

~~DO 230 J=1,N~~
 230 COPI(I,J-1)=A(I,J)

00168*22
 00169*13

~~GO TO 224~~

222 CONTINUE

~~DO 223 J=1,NPK~~
 223 COPI(I,J)=A(1+2,J)

224 CONTINUE

220 CONTINUE

00170*13

JUBN=10
 CALL EIGRS(COPI,NPK,JUBN,KW,XP,NPK,SG,IER)
 255 WRITE(6,255) (KW(I),I=1,NPK)
 FURMA1(5X,12E10.3)
 WRITE(6,256) IER

253	DO 253 I=1,MM DO 253 J=1,MM C01(I,J)=0.0	
257	DO 257 I=1,MM S0(I)=0.0	
	TOL=0.100E-32	
288	DO 242 I=1,NPK IF(RW(I)-TOL) 288,289,289 CONTINUE WRITE(6,250)I,RW(I),TOL STOP	00178*19 00180*15
289	CONTINUE	
242	CONTINUE	
		00175*13
		00177*15
250	FORMAT(// ' MATRIX A IS NEGATIVE SEMIDEFINITE ' * ' ELEMENT NO. = ' * ' EIGENVALUE(1) = ' * ' TOL, = ',E10.3/ = ',E10.3/)	00179*15
290	CONTINUE	00181*15
	WRITE(6,260)	00182*19
260	FORMAT(// ' MATRIX A IS POSITIVE SEMIDEFINITE ' GO TO 275	00183*15 00184*20
295	CONTINUE	00185*15
	WRITE(6,270)	00186*19
270	FORMAT(// ' MATRIX A IS POSITIVE DEFINITE ' CONTINUE	00187*15 00188*13 00189*13 00190*13 00191*19
275	CONTINUE	
	IF(KPT.EQ.1.AND.NPT.EQ.0) GO TO 495	
455	DO 455 I=1,N CE(I)=0.0 CE(I)=1.0	00192*13 00193*19 00194*19 00195*21 00196*21
451	DO 451 I=1,NPK A(I,1)=AM(I)	00197*13 00198*13 00199*23
	DO 460 I=1,NPK DO 450 J=2,N	00200*23 00201*23 00202*23
450	A(I,J)=-A(I,J)	00203*23
460	CONTINUE	00204*23 00205*32 00206*32 00207*34

GO TO 490 00208*34
 ***** LINKED HERE THE BRANCHING OF RESIDUAL STRESSS OP00209*32
 ***** 00210*32

480 CONTINUE 00211*32
 00212*33

NPP=NPK+1 00213*32
 NPU=NPP+1 00214*32
 NPD=N 00215*48
 DU 457 I=1,NPK 00216*32
 DU 462 J=2,NPP 00217*33
 462 A(I,J)=AH(J-1,I) 00218*33
 DU 456 J=NPH,NPD 00219*32
 456 A(I,J)=-AN(J-NPP,I) 00220*32
 457 CONTINUE 00221*32
 00222*48
 00223*48
 00224*48
 00225*48
 00226*32

NH=NPK+1 00227*32
 NH1=NPK+NEWS 00228*32
 DU 458 I=NH,NH1 00229*32
 DU 459 J=2,NPP 00230*49
 459 A(I,J)=CT(J-1,I-NPK) 00231*34
 00232*34

DU 465 J=NPU,NPD 00233*32
 465 A(I,J)=-CI(J-NPP,I-NPK) 00234*49
 00235*48
 00236*34
 00237*34
 458 CONTINUE 00238*35
 ***** PRINT OF MATRICES BY PARIS ***** 00239*34
 00240*34

DU 485 I=1,N 00241*36
 485 CE(I)=0.0 00242*36
 CE(I)=1.0 00243*36
 00244*36
 00245*36
 00246*36

IF(NPT.EQ.2) GO TO 488 00247*45
 00248*44
 00249*36
 00250*36

DU 467 I=1,NPK 00251*36
 487 A(I,I)=AH(I) 00252*34
 00253*32
 00254*32
 72 FURNAL(DX,12E10,J) 00255*32
 00256*34
 00257*34
 00258*34

490 CONTINUE 00259*35
 DU 452 I=1,NPK 00260*13
 452 SQ(I)=FK(I) 00261*13
 00262*44
 GO TO 495 00263*44

```

488 CONTINUE
NPE=NPK+1
NPG=NPK+NEQS
DO 487 I=NPE,NPG
489 A(1,I)=-GB(1-NPK)
DO 498 I=1,NPK
498 SU(1)=PK(1)
DO 499 I=NPE,NPG
499 SU(1)=SU(1-NPK)

495 CONTINUE

WRITE(6,195)
195 FORMAT(1H1/' **** PRINT OF VECTOR MB **** ')
WRITE(6,196) (GB(1),I=1,NEQS)
196 FORMAT(5X,12E10.3)
WRITE(6,197)
197 FORMAT(1H1/' ** PRINT OF VECTOR SU ** ')
WRITE(6,198) (SU(1),I=1,NEQS)
WRITE(6,198)
198 FORMAT(1H1' *** PRINT OF DIMENSION FOR OPTIMIZATION *** ')
WRITE(6,199) NPC,N,M1,M2
199 FORMAT(5X,415)

IF(NPK.NE.0) GO TO 10000
WRITE(6,9990)
9990 FORMAT(1H1/' *** PRINT OF N-VECTOR **** ')
WRITE(6,9991) (A(1,I),I=1,NPK)
9991 FORMAT(1X,12E10.3)
WRITE(6,9992)
9992 FORMAT(1H1/' *** PRINT OF N-TRANSPOSE *** ')
WRITE(6,9993) ((A(1,J),J=2,NPK),I=1,NPK)
WRITE(6,9993)
9993 FORMAT(1H1/' **** PRINT OF -N - TRANSPOSE *** ')
WRITE(6,9994) ((A(1,J),J=NPK,NPD),I=1,NPK)
WRITE(6,9994)
9994 FORMAT(1H1/' **** PRINT OF N - TRANSPOSE *** ')
WRITE(6,9995) ((A(1,J),J=2,NPP),I=NH,NH1)
WRITE(6,9995)
9995 FORMAT(1H1/' **** PRINT OF -C - TRANSPOSE *** ')
WRITE(6,9996) ((A(1,J),J=NPD,NPD),I=NH,NH1)
WRITE(6,9996)
9996 FORMAT(1H1/' *** PRINT OF LOAD VECTOR *** ')
WRITE(6,9997) (A(1,I),I=NH,NH1)
WRITE(6,9998)
9998 FORMAT(1H1/' **** PRINT OF K VECTOR ** ')
WRITE(6,9999) (SU(1),I=1,NPG)
WRITE(6,9999)

```

```

00264*44
00265*44
00266*44
00267*44
00268*44
00269*44
00270*44
00271*44
00272*54
00273*44
00274*44
00275*44
00276*44
00277*50
00278*44
00279*44
00280*44
00281*44
00282*34
00283*34
00284*51
00285*51
00286*51
00287*51
00288*54
00289*55
00290*51
00291*51
00292*52
00293*51
00294*51
00295*51
00296*51
00297*56
00298*56
00299*56
00300*56
00301*56
00302*56
00303*56
00304*56
00305*56
00306*56
00307*56
00308*56
00309*56
00310*56
00311*56
00312*56
00313*56
00314*56
00315*56
00316*56
00317*56
00318*56
00319*56
00320*56
00321*56
00322*56
00323*56

```

00324*56
00325*56

*** PRINT OF OBJECTIVE FUNCTION *** ')

9999 FORMAT(IHI) (CE(I),I=1,N)
WRITE(6,9991) (CE(I),I=1,N)

```

0000 CONTINUE.                                00326*56
                                                00327*56
                                                00328*56
GO TO(80,70),KPT
70 CONTINUE
***** CALL OPTIMIZATION ROUTINE *****
CALL ZA3LP(A,LPC,SQ,CE,N,M1,M2,SG,PSOL,DSOL,KW,IW,IER)
***** TEST FOR SEMIDEFINITENESS
190 FORMAT(// ' MATRIX Z (INFLUENCE) IS NOT NEG SEMIDEFINITE ')
PR=0.0
DO 191 I=1,NPK
191 PR=PR+DSOL(I)*SQ(I)
WRITE(6,192)
192 FORMAT('192' CHECK BY MINIMIZATION '/')
WRITE(6,193) PR,(DSOL(I),I=1,NPK)
193 FORMAT(5X,12E10.3)
GO TO 90
00329*56
00330*13
00331*13
00332*13
00333*19
00334*31
00335*31
00336*40
00337
00338
00339*13
00340*21
00341*21
80 CONTINUE
IF(NPT.EQ.0) GO TO 85
DO 86 I=1,NEQ
DO 87 J=1,NPK
87 A((I+2),(J))=AN(I,J)
DO 88 K=1,NEQS
A((I+2),(NPK+K))=-CT(I,K)
A((I+2),(K+NPK+NEQS))=C1(I,K)
88 CONTINUE
86 CONTINUE
85 CONTINUE
CALL ZXMIN(CE,A,AN,C,AM,FK,GB,SU,SN,ICOLMS,IDES,ROW,CUPI,PSOL,
*DSOL,KW,IER,NPK,NPJ,NEQS,NUMEL,NYP,NSC,SD,SN,CT,IH,NUMNP)
GO TO 95
00342*28
00343*28
90 CONTINUE

```

CALL OUTPUT(NUMEL,NYP,NSC,NPJ,NPK,PSOL,SG,AN,SD,SN,CT,SU,NEGS,IH,
*NUMNP) 00344*46

95 CONTINUE

00345*13

00346

00347

00348*28

00349*23

00350*23

00351*29

00352*29

00353*38

00354*39

00355*38

00356*29

00357*29

00358*23

00359*23

00360*29

00361*54

00362*23

00363*23

00364*24

00365

00366

IF(KK.EQ.0) GO TO 1300
CONTINUE

1201

***** READ INFORMATION ABOUT FOR WHICH BOUNDING TO B

READ(5,1400) NDIY,NNOD

READ(5,1400) ((JK(I,J),J=1,3),I=1,NNOD)

1400

FORMAT(5X,815)

***** PRINT ALL INFORMATION ABOUT NODE TO BE BUUN

CALL BOUND(NEGS,NPJ,NPK,SN,AS,C,SP,S,SD,SG,SQ,CE,PSOL,EK,A,M2,RW,

*IW,LPC,DSOL,AN,MN,IH,PPSOL,DDSOL,SH,XP)

1300 CONTINUE

RETURN

END

00053
00054
00055
00056
00057
00058**3
00059**3
00061**3

A(I)=0.
B(I)=0.
C(I)=0.
D(I)=0.
10 B(I)=0.

***** ADD INITIALIZATION TO DYNAMIC ANALYSIS COMPLETED*****

IF(KKS.NE.4) GO TO 400
IF(NDYN.NE.1) GO TO 400
IF(CMMS.EQ.1) GO TO 400

***** READ HERE PREVIOUS STORED DISPL. VELOCITY COMPONENTS

READ(58,400) (A(I),I=1,NEU)
READ(59,400) (X1(I),I=1,NEU)

405 FORMAT(5A,3I5)
406 FORMAT(5X,12F10.3)

400 CONTINUE
RETURN
END

00062
00063


```

DATA SET INPUT AT LEVEL 005 AS OF 81/11/03
SUBROUTINE INPUT (ID,K,Z,IH,NUMNP,NEQ,NEQH,NEQS) 00001
IMPLICIT REAL*8 (A-H,O-Z) 00002
00003
NUDAL POINT INPUT AND GENERATION. 00004
00005
*****CHANGED BECAUSE OF NEW ARRAY CREATED FOR SHAKEDOWN ANALYS 00006
DIMENSION K(1),Z(1),ID(NUMNP,4),NID(4),IH(1) 00007
DIMENSION R(1),Z(1),ID(NUMNP,4),NID(4),IH(NUMNP,2) 00008
REWIND 48
REWIND 49
ND=3 00009
WRITE(6,200) 00010
WRITE(6,204) 00011
KU=1 00012
DO 5 I=1,4 00013
5 NID(I)=0 00014
11 READ(5,100) N,(ID(N,1),I=1,4),R(N),Z(N),KN 00015
WRITE(6,203)N,(ID(N,1),I=1,4),R(N),Z(N),KN 00016
DO 60 I=1,4 00017
IF (ID(N,I)) 01,62,63 00018
61 NID(I)=-1 00019
ID(N,I)=1 00020
GO TO 60 00021
62 IF (NID(I).EQ.-1) ID(N,I)=1 00022
GO TO 60 00023
63 NID(I)=0 00024
60 CONTINUE 00025
IF (ND.EQ.1) GO TO 12 00026
CHECK IF GENERATION NEEDED 00027
IF (KN) 10,10,20 00028
12 KU=0 00030
10 CONTINUE 00031
NUMINT=1 00032
GO TO 15 00033
00034
GENERATION NEW NODES 00035
20 NUMINT=(N-N1)/KN 00036
DK=(R(N)-R(N1))/NUMINT 00037
DZ=(Z(N)-Z(N1))/NUMINT 00038
NUMINT=NUMINT-1 00039
DO 21 J=1,NUMINT 00040
NN=N1+J*KN 00041
R(NN)=R(NN-KN)+DK 00042
Z(NN)=Z(NN-KN)+DZ 00043
00044
00045
SET BOUNDARY CODES ... SAME AS FIRST JOINT IN SERIES 00046
DO 22 JJ=1,4 00047
IF (ID(N1,JJ)-1) 24,26,25 00048
00049
00050
GENERATE NEW MASTER NODES 00051
25 ID(NN,JJ)=ID(N1,JJ)+J*KN 00052
GO TO 22 00053
26 ID(NN,JJ)=ID(N1,JJ) 00054
GO TO 22 00055
24 ID(NN,JJ)=0 00056
22 CONTINUE 00057

```

```

21 CONTINUE                                00058
15 N1=N                                    00059
                                           00060
CHECK FOR LAST NODAL POINTS                00061
                                           00062
IF(NUMNP-N1) 13,13,11                      00063
13 CONTINUE                                 00064
                                           00065
PRINT ALL NODAL DATA                       00066
                                           00067
WRITE(6,202)                                00068
WRITE(6,204)                                00069
DO 50 N=1,NUMNP                             00070
                                           00071
WRITE(6,203) N,(ID(N,1),I=1,4),R(N),Z(N)    00071
WRITE(6,203) N,(ID(N,1),I=1,4),R(N),Z(N)    00071
50 CONTINUE                                  00072
*****ARRAY IH(NUMNP,2) IS BEING CREATED HERE. IT WILL BE THE 00073
*****CALLED IN ELSIF ROUTINE FOR COMPUTING THE LM ARRAY 00074
NEQS=0                                       00075
DO 750 I=1,NUMNP                            00076
DO 700 J=1,2                                 00077
IF(ID(I,J)) 680,650,680                     00078
650 NEQS=NEQS+1                              00079
IH(I,J)=NEQS                                 00080
GO TO 700                                    00081
680 IH(I,J)=-1                               00082
700 CONTINUE                                 00083
750 CONTINUE                                 00084
*** ARRAY IH(NUMNP,2) IS STORED IN NEW LOCATIONS ( N4 TO N5) 00085
*****PRINT HERE IN ARRAY AND NEQS 00086
*****FINISH PRINTING 00087
NEQ=0                                       00088
NEQH=0                                       00089
DO 75 I=1,NUMNP                             00090
DO 70 J=1,4                                  00091
IF(ID(I,J)) 68,65,68                        00092
65 NEQ=NEQ+1                                 00093
ID(I,J)=NEQ                                  00094
GO TO 70                                     00095
68 ID(I,J)=-1                               00096
70 CONTINUE                                 00097
75 CONTINUE                                 00098
NEQH=NEQ                                     00099
WRITE(6,205) NEQ,NEQH                      00100
                                           00101
RETURN                                       00102
                                           00103
100 FORMAT (5I5,2F10.4,15)                 00104
104 FORMAT (15,4F10.4)                     00105
200 FORMAT(1H1//20H1NODAL POINT DATA AS INPUT//) 00106**4
202 FORMAT(1H1//20H1COMPLETE NODAL POINT DATA // ) 00107**4
203 FORMAT (15,4I5,2F10.4,15)              00108
204 FORMAT (5H3NODE 3X 10H B.C.CODES 11X   00109
* 23H1NODAL POINT COORDINATES / 7H NUMBER 2X, 1HX,4X,1HY 4X 2HWX 3X 00110
* 2HWY 11X 1HX 12X 1HY/)                  00111
205 FORMAT(1H1/30H NUMBER OF EQUATIONS ..... I10/ 00112**5
* 30H NUMBER OF ROWS OF H 110)           00113
                                           00114

```

DATA SET INSDMA AT LEVEL 002 AS UF 81/09/04
 SUBROUTINE LUSYMA(MP,U,A)
 IMPLICIT REAL*8(A-H,O-Z)

INVERSE SYMMETRIC MATRIX

DIMENSION A(MP,M,MP)

DO 200 N=1,MP

DIAG=A(N,N)

DO 100 J=1,MP

100 A(N,J)=-A(N,J)/DIAG

DO 150 I=1,MP

IF(N.EQ.I) GO TO 150

DO 140 J=1,MP

IF(N.EQ.J) GO TO 140

120 CONTINUE

A(I,J)=A(I,J)+A(I,N)*A(N,J)

140 CONTINUE

150 A(I,N)=A(I,N)/DIAG

A(N,N)=1.0/DIAG

200 CONTINUE

RETURN

END

00001
 00002
 00003
 00004
 00005
 00006
 00007
 00008
 00009
 00010**2
 00011
 00012
 00013**2
 00014**2
 00015
 00016
 00017
 00018
 00019
 00020
 00021

DATA SET ITRSLD AT LEVEL 010 AS OF 83/02/02
 SUBROUTINE ITRSLD(NEL,I,GSTRES,FACN,PS,IL,PG,LM,XP)

IMPLICIT REAL *8(A-H,O-Z)

DIMENSION GSTRES(9),FACN(4),PS(16,8),P(4,8),XP(16),IS(8)

*,IL(8),LM(16),FI(5,12)

DATA IS/1,2,5,6,9,10,13,14 /,IL/3,4,7,8,11,12,15,16/

TAU11=GSTRES(1)*FACN(L)
 TAU22=GSTRES(2)*FACN(L)
 TAU33=GSTRES(3)*FACN(L)
 TAU12=GSTRES(4)*FACN(L)
 TAU=PG+FACN(L)

DO 300 I=1,8
 DO 300 JJ=1,4
 J=JJ+*(I-1)
 P(JJ,1)=PS(J,I)

300 CONTINUE

301 WRITE(6,301)
 FORMAT(1X,' PRINT OF STRAIN MATRIX AND VOLUME ')
 WRITE(6,302) ((P(I,J),J=1,8),I=1,4)
 WRITE(6,302) (FACN(I))
 302 FORMAT(5X,8E10.3)

DO 310 II=1,8
 NS=15(II)
 NF=16(II)
 XP(NS)=XP(NS)+P(1,II)*TAU11+P(2,II)*TAU22+P(3,II)*TAU33+P(4,II)*
 *TAU12
 XP(NF)=XP(NF)+FI(1,II)*TAU

310 CONTINUE

**** WRITE INFORMATION ON TAPE FOR ASSEMBLING ****

160 WRITE(6,160)
 FORMAT(5X,' *** PRINT IN ITRSLD ***',/)
 WRITE(6,161) (XP(I),I=1,16)
 161 FORMAT(5X,8E10.3)

RETURN
 END

00001**5
 00002**5
 00003**2
 00004**2
 00005**2
 00006**7
 00007**7
 00008**2
 00009**2
 00010**3
 00011**2
 00012**2
 00013**8
 00014**2
 00015**2
 00016**2
 00017**2
 00018**2
 00019**2
 00020**2
 00021**7
 00022**2
 00023**6
 00024**7
 00025**7
 00026**2

00027**2
 00028**2
 00029**2
 00030**2
 00031**2
 00032**2
 00033**2
 00034**2
 00035**2
 00036**2
 00037**2

00038
 00039


```

C2=1.0/(BETA*H1*D1)
C3=GAMA/(BETA*D1)
REWIND 13
00060
00061
00062
00063
00064
00065
00066
00067
00068
00069
00070*22
00071*22
00072*22
00073*22

REWIND 9
REWIND 4

DO 900 N=1,NUMEL
*****CHANGED FROM ORIGINAL FOR STATIC TRANSIENT PROBLEM****
00074*22
00075
00076
00077
00078
00079
00080
00081
00082*27
00083*27
00084*27
00085*27
00086*27
00087*27
00088*27
00089*29
00090*29
00091*19
00092
00093
00094
00095
00096
00097**2
00098
00099
00100
00101
00102

READ(13) LM,S,EM,H,SS,XP,XN
WRITE(6,39)
39 FORMAT(5X,' PRINT IN KSTAR DAMPING MATRIX ')
WRITE(6,40) ((H(I,J),J=1,16),I=1,16)
40 FORMAT(2X,16E8.1)

DO 360 I=1,16
L=LM(I)
GO TO(400,400,400,500,500),NDYN
400 IF(L) 360,360,200

200 CONTINUE
IF(NDYN.EQ.4.OR.NDYN.EQ.5) GO TO 201

XN(L)=XN(L)+EM(I)

201 CONTINUE
***** CHECK FOR RESIDUAL LOAD VECTOR *****
RE(L)=RE(L)+XP(I)
SW(L)=SW(L)+XN(I)
500 DO 350 J=1,16
KL=LM(J)
IF(KL) 350,350,250
250 IF(KL-L) 350,300,300
300 K=KL-L+1
A3(L,K)=A3(L,K)+SS(I,J)
A4(L,K)=A4(L,K)+S(I,J)
A5(L,K)=A5(L,K)+H(I,J)
350 CONTINUE
360 CONTINUE
900 CONTINUE
WRITE(6,37)
37 FORMAT(1H1/5X,' PRINT IN KSTAR ')
WRITE(6,38) (XN(I),I=1,NEQ)
38 FORMAT(5X,12E10.3)
WRITE(6,36)((A3(I,J),J=1,MBAND),I=1,NEQ)
WRITE(6,36)((A5(I,J),J=1,MBAND),I=1,NEQ)

IF(KKS.NE.3) GO TO 16
IF(NC.EQ.0) GO TO 16
00103*35
00104*35
00105*34
00107*36

15 LCOUNT=LCOUNT+1
WRITE(6,18) LCOUNT,1STRPR

```

18	FORMAT(5X,215)	00108*34
	IF(LDCOUNT.NE.1SIRFR) GO TO 20	00109*34
		00110*34
		00112*34
		00113
16	CONTINUE	
	WRITE(6,34)	00114*15
34	FORMAT(1H1/' *** PRINT OF RESIDUAL LOAD VECTOR **** ')	00115*15
	WRITE(6,36) (RE(1),I=1,NEQ)	00116*15
	WRITE(6,35)	00117*20
35	FORMAT(1H1/' *** PRINT OF SELF LOAD VECTOR ***** ')	00118*20
	WRITE(6,36) (SW(1),I=1,NEQ)	00119*20
36	FORMAT(5X,8E12.5)	00120*15
		00121
		00122*17
		00123
		00124*18
		00125
		00126
	***** CHECK REWIND LOCATION IN MAIN *****8	00127*31
	REWIND 00	00128*31
		00129*31
999	FORMAT(//)	00130*31
		00131*31
		00132*29
		00133
		00134
		00135*33
		00136
		00137*26
970	FORMAT(1H1/' ***** PRINT OF ASSEMBLED MATRIX IN KSTAR **')	00138*26
	WRITE(6,971) ((A4(I,J),J=1,MBAND),I=1,NEQ)	00139*26
	PRINT 999	00140*26
	WRITE(6,972)	00141*29
972	FORMAT(5X,' PRINT OF MASS MATRIX ')	00142*29
	WRITE(6,971) (AM(I),I=1,NEQ)	00143*29
	WRITE(6,971) ((A4(I,J),J=1,MBAND),I=1,NEQ)	00144*30
800	CONTINUE	00140*26
		00145*29
		00146*26
	WRITE(60) ((A4(I,J),J=1,MBAND),I=1,NEQ)	00147*28
		00148*28
		00149*28
		00150*34
20	CONTINUE	00151*34
971	FORMAT(5X,10E10.3)	00152*32
		00153*29
		00154*26
		00155
	WRITE(4) ((A4(I,J),I=1,NEQ),J=1,MBAND)	00156*17
	WRITE(9) (AM(I),I=1,NEQ)	00157
	NEQB=NEQ	00158
	MA=MBAND	00159
	NBBLOCK=1	00160
	NEIG=0	00161
	NWA=NEQB*MBAND	00162
	NIB=(MBAND-2)/NEQB+1	00163
	IF(NIB.GE.NBLOCK) NIB=NBLOCK-1	00164

```

MI=NEQB+NBAND-1
00165
00166*23
IF(NDYN.EQ.5) GO TO 960
00167*23
00168*23
00169*23
00170*23
00171*23
00172*23
00173*23
00174*23
00175
*****00176
IF(NDYN.EQ.4) GO TO 1000
00177
DU 961 I=1,NEQ
00178
00179
00180
00181
00182
00183
IF(NDYN.EQ.5) GO TO 9610
00184
A4(I,1)=A4(1,1)+A0*XM(I)+A02*A5(1,1)
00185
GO TO 961
00186
9610 A4(I,1)=A4(1,1)+(1./D11)*A5(1,1)
00187
961 CONTINUE
00188
DU 962 J=2,NBAND
00189
DU 962 I=1,NEQ
00190
00191
00192
00193
00194
00195
IF(NDYN.EQ.5) GO TO 9620
00196
A4(I,J)=A4(1,J)+A02*A5(1,J)
00197
GO TO 962
00198
9620 A4(I,J)=A4(1,J)+(1./D11)*A5(1,J)
00199
962 CONTINUE
00200
00201
00202
00203
00204
INFORMATION FOR ENTERING IN TRIFAC MATRIX DECOMPOSITION ***00205
1000 CONTINUE
00206
REWIND 4
00207
WRITE(4) ((A4(1,J),I=1,NEQ),J=1,NBAND)
00208
CALL TRIFAC(A4,A6,MAXA,NEQB,NA,NBLCK,NWA,NTB,NEQ,MI)
00209
4000 CONTINUE
00210
00211
00212
00213
1001 CONTINUE
00214
00215
00216
RETURN
00217
END
00218

```



```

DATA SET LOAD          AT LEVEL 027 AS OF 83/02/16
SUBROUTINE LOAD(INEL,PRESS,VOL,WA)
IMPLICIT REAL*8(A-H,O-Z)
COMMON/CONT1/DT,HED(8),KADN,NASIN,NOHEL,NEG,MBAND,NCYCL,NDYN,IPLNA
*X,ISTKPK,NPRINT,ISTART,NEQH,MEANDH
COMMON/EMB/ S(16,16),GM(16),GN(16,16),SS(16,16),XP(16),XN(
*16),LM(16)
***** ELEMENT LOAD VECTOR EVALUATED FROM ELEMENT CENTROID
COMMON/BFLA/NU,NELTYP,ND,NUMMAT,MODEL,NCUN,LDW,NK015,NK016,NK017
COMMON/ NANCY/ CM(12000)
COMMON/BOBBY/KKS,NUMGAT,NC,LOOP,NDRN,NINT,LDWA,NPT
COMMON/QUAD/RK(4),ZZ(4),FAC,R,U(6),P(4,12),SF(12,12),H(12,12),
*ST(20,12),D(4,4),C(12,12),E(12,12),TT(5,12),PI(4,8)
COMMON/GURA/IFLAG,ITR
COMMON/RAH/RM(5),ZM(5),IPT
DIMENSION IS(8),IH(8)
DATA IS/1,2,5,6,9,10,13,14/,IH/3,4,7,8,11,12,15,16/
DIMENSION WA(10WA,1),PRESS(5)
GO TO (400,400,500,400),KKS
500 IF (IFLAG.EQ.0) GO TO 400
NG=15*(INEL-1)*NPT+(IPT-1)*15
DO 200 J=1,5
200 PRESS(J)=CM(NG+10+J)
400 CONTINUE
TAU11=PRESS(1)*FAC
TAU22=PRESS(2)*FAC
TAU33=PRESS(3)*FAC

```

```

00001**6
00002**5
00003
00004**5
00005
00006
00007*21
00008*21
00009**2
00010**2
00011**2
00012
00013*27
00014**9
00015**9
00016**9
00017*15
00018
00019
00020**2
00021**2
00022*18
00023*18
00024*26
00025*23
00026*25
00027**2
00028**2
00029**2
00030**2
00031**2
00032**2
00033*15
00034**9
00035*21
00036*23
00037*23
00038*23
00039*23
00040*21
00041**9
00042*15
00043*15
00044**9
00045*19
00046**9
00047*22
00048*22
00049*22
00050**9
00051**9
00052*15
00053*15
00054*16
00055*15
00056*15
00057*15
00058*15
00059*15

```

```

TAU12=PRESS(4)*FAC
DO 300 11=1,6
NS=IS(11)
NF=IL(11)
XP(NS)=XP(NS)+P(1,11)*TAU11+P(2,11)*TAU22+P(3,11)*TAU33+P(4,11)
**TAU12
XP(NF)=XP(NF)+11(1,11)*PRESS(5)*FAC
300 CONTINUE
999 FORMAT(//)
510 FORMAT(1X,6E12.5)
RETURN
END

```

```

00060*15
00061*15
00062*15
00063*15
00064*15
00065*15
00066*15
00067*17
00068*17
00069*17
00070*15
00071*15
00072*15
00073*15
00074*15
00075**9
00076*12
00077*12
00078*12
00079*20
00080*20
00081*14
00082**9
00083
00084*24
00085*24
00086

```

```

DATA SET LOAD1      AT LEVEL 035 AS OF 83/03/05
SUBROUTINE LOAD1(NUMNP,NEQ,IC,ISLC,PI,NC,ISC,JSC,SURTRX,SURTRY, 00001
*X,Y,0,10,A01,IK,IM,PDIL,SURPTA,SURPFY) 00002*10
IMPLICIT REAL*8(A-H,O-Z) 00003**2
00004*10
00005**6
00006**6
00007**6
COMMON/ASIM/NIP,NSC,NSLC,NLC,NNYP,NNSC 00008*18
00009*18
COMMON/BOB/ANUM,NG,LOCK,NEQ,NDC
00010*18
COMMON/KRISNA/SCALB(50),SCALF(50),LL 00011*30
00012*30
COMMON/BOBBY/KKS,NUMMAT,NG,LOOP,NDRN,NINT,IDWA,NPT 00013*32
00014*32
COMMON/PETER/AREA,WIDTH,HOR,VERT,ACCG 00015*31
00016*31
DIMENSION Q(1),X(1),Y(1),ID(2),ISC(1),JSC(1),SURTRX(2,NDC,2), 00017*20
*SURTRY(2,NDC,2),ID(NUMNP,1),TS(2),PDYL(2,4,NUMNP) 00018*20
DIMENSION SURPFX(2,NDC,2),SURPFY(2,NDC,2) 00019*20
00020
***** TEMPORARY STORAGE ***** 00021
DIMENSION AU1(1) 00022
1STOP=0 00023**8
IF(NSLC.EQ.0) GO TO 900 00024
00025**5
00026
00027
00028
***** SURFACE LOAD PRESCRIBED HERE ** STATIC OR 00029
***** CHECK VARIABLE (1) ***** 00030
NJ=NUMNP*4 00031
DO 2 I=1,NJ 00032
2 Q(I)=0.0 00033
00034
00035
00036
***** CONDITION CHECK ***** 00037
IF(NC.NE.0) GO TO 700 00038
00039
READ AND PRINT SURFACE LOADING (TRACTION) CARDS 00040
30 WRITE(6,108)LC 00041
KL=1 00042
55 CONTINUE
00043
DO 40 N=KL,2 00044
READ(5,1012) ID(N) 00045
IF(NC.GT.13) GO TO 57
WRITE(6,2016) ID(N) 00046
57 CONTINUE 00047
DO 40 L=1,NSLC 00048**8
00049**8
READ(5,41)ISC(L),JSC(L),SURTRX(N,L,1),SURTRX(N,L,2),SURTRY(N,L,1), 00050
*SURTRY(N,L,2) 00051*27
00052*29

```

```

SURTRX(N,L,1)=SURTRX(N,L,1)*SCALB(L)      00053*29
SURIRX(N,L,2)=SURIRX(N,L,2)*SCALB(L)      00054*29
SURTRY(N,L,1)=SURTRY(N,L,1)*SCALB(L)      00055*29
SURTRY(N,L,2)=SURTRY(N,L,2)*SCALB(L)      00056*29
                                           00057*29
                                           00058*27
                                           00059*28
                                           00060*27
IF(NDRN.FU,0) GO TO 59                      00061*27
***** READ FLUID LOADING *****           00062*27
READ(5,41) ISC(L),JSC(L),SURPFX(N,L,1),SURPFX(N,L,2),SURPFY(N,L,1) 00063*27
*,SURPFY(N,L,2)                             00064*27
                                           00065*29
SURPFX(N,L,1)=SURPFX(N,L,1)*SCALF(L)      00066*29
SURPFX(N,L,2)=SURPFX(N,L,2)*SCALF(L)      00067*29
SURPFY(N,L,1)=SURPFY(N,L,1)*SCALF(L)      00068*29
SURPFY(N,L,2)=SURPFY(N,L,2)*SCALF(L)      00069*29
                                           00070**8
59 CONTINUE                                00071**8
                                           00072
IF(NC.GT.13) GO TO 58                      00072
WRITE(6,42)                                 00072
*ISC(L),JSC(L),SURTRX(N,L,1),SURTRX(N,L,2),SURTRY(N,L,1),
1 SURTRY(N,L,2),SURPFX(N,L,1),SURPFX(N,L,2),SURPFY(N,L,1),SURPFY(N,
*L,2)                                       00073**3
                                           00074**3
                                           00075**8
58 CONTINUE                                00076**8
40 CONTINUE                                00077**8
108 FORMAT(54HINPUT TABLE 5.. SURFACE LOADING DATA FOR LOADING CASE,
*15//)                                       00078
                                           00079
2016 FORMAT(                                00080
117X, 43HSURFACE LOAD INTENSITIES AT NODES FOR TIME,F10.7/
24X,6HNODE I,4X,6HNODE J,10X,2HXI,10X,2HXJ,10X,2HYI,10X,2HYJ) 00081
                                           00082
41 FORMAT(214,4E11.4)                       00083*27
42 FORMAT(2110,8F12.5)                     00084**3
CONVERT LINEARLY VARYING SURFACE TRACTIONS TO STATIC EQUIVALENTS,
AND ADD TO OVERALL LOAD VECTOR R, EQ.(5-61A). 00085
***** SET UP LOAD MATRIX                 00086
                                           00087
                                           00088*12
                                           00089*12
                                           00090*12
IF(XL.EQ.2) GO TO 117                       00091
KL=2                                         00092
TM=TD(1)+DT                                  00093
L=2                                           00094
DTXX=TD(2)-TD(1)                            00095
                                           00096*12
                                           00097*12
700 CONTINUE                                00098
IF(TM.GT.TK) GO TO 120                      00099
                                           00100*35
                                           00101*33
                                           00102*33
3 CONTINUE                                  00103*34
                                           00104*34
WRITE(6,1) DT,NC,FM,TK,TD(2),TD(1)         00105*33
1 FORMAT(5X,1 DT = 1,F5.2/
* 5X,1 NC = 1,15/

```



```

FXI=SLOPE5*EI
FXJ=SLOPE6*EI
FYI=SLOPE7*EI
FYJ=SLOPE8*EI
GU TO 180
135 SLOPE1=SURTRX(2,L,1)
SLOPE2=SURTRX(2,L,2)
SLOPE3=SURTRY(2,L,1)
SLOPE4=SURTRY(2,L,2)
SLOPE5=SURPFX(2,L,1)
SLOPE6=SURPFX(2,L,2)
SLOPE7=SURTFY(2,L,1)
SLOPE8=SURTFY(2,L,2)

```

```

TME=1.0
PXI=SLOPE1*EI*TME
PXJ=SLOPE2*EI*TME
PYI=SLOPE3*EI*TME
PYJ=SLOPE4*EI*TME
FXI=SLOPE5*EI*TME
FXJ=SLOPE6*EI*TME
FYI=SLOPE7*EI*TME
FYJ=SLOPE8*EI*TME
180 CONTINUE

```

```

***** LINEAR VARIATION LOAD CALCULATION ON NODES **
Q(I1-3)=Q(I1-3)+PXI/3.0+PXJ/6.0
Q(JJ-3)=Q(JJ-3)+PXI/6.0+PXJ/3.0

```

75 CONTINUE.

76 CONTINUE

```

***** BRANCH AT THIS POINT FOR PARABOLIC SHEAR LOADING *****
Q(I1-2)=Q(I1-2)+PYI/3.0+FYJ/6.0
Q(JJ-2)=Q(JJ-2)+PYI/6.0+FYJ/3.0
Q(I1-1)=Q(I1-1)+FXI/3.0+FXJ/6.0
Q(JJ-1)=Q(JJ-1)+FXI/6.0+FXJ/3.0
Q(I1)=Q(I1)+FYI/3.0+FYJ/6.0
Q(JJ)=Q(JJ)+FYI/6.0+FYJ/3.0

```

```

***** CALCULATE ALSO BY PARABOLIC METHOD *****
130 CONTINUE
DO 600 M=1,NUMNP
JP=J

```

```

00168**3
00169**3
00170**3
00171**3
00172
00173
00174
00175
00176
00177**3
00178**3
00179**3
00180**3
00181**8
00182**8
00183**8
00184
00185
00186
00187
00188
00189**3
00190**3
00191**3
00192**3
00193
00194**8
00195**8
00196**8
00197*15
00198**8
00199**8
00200**8
00201
00202
00203*15
00204*15
00205*15
00206*15
00207*15
00208*13
00209*13
00210*13
00211*13
00212*13
00213
00214
00215**3
00216**3
00217**3
00218**3
00219**8
00220**8
00221**8
00222**8
00223**8
00224**8
00225
00226
00227

```

DU 600 J=1,4	00228
KK=4*M-JP	00229
PDYL(Z,J,M)=Q(KK)	00230
XX=X(1)	
IF(1P1MAX.NE.1) GO TO 601	
PDYL(Z,J,M)=Q(KK)*(XX)	
601 CONTINUE	
JP=JP-1	00231
600 CONTINUE	00232
	00233**8
	00234**8
	00235**8
	00236**8
DU 140 I=1,NUMNT	00237
DU 140 J=1,4	00238
JJ=1D(1,J)	00239
IF(JJ) 140,140,150	00240
150 A01(JJ)=PDYB(Z,J,I)	00241
140 CONTINUE	00242
	00243**8
	00244**8
900 CONTINUE	00245*24
RETURN	00246
1012 FORMAT(8F10.0)	00247
2010 FORMAT(/'ERROR IN BASE ACCEL OR DIN LOAD,TIME = ',F10.3)	00248
END	00249

```

DATA SET LOADZ      AT LEVEL 009 AS OF 83/02/11
SUBROUTINE LOADZ(NUMNP, PDYL, X, NUMLP, ID, DT, TD, NC, TK, TM, AU2)
IMPLICIT REAL*8 (A-H,O-Z)

DIMENSION PDYL(2,4,1),TD(2),K(1),ID(NUMNP,1)
COMMON/CONIL/DD,HED(8),RADN,NASIH,NUMSL,NKK,MBAND,NCYCL,NDYN,IPLNA
*X,1STRPR,NPRINI,1START,NEQK,MBANDH
*,NEQS,MBANDS

COMMON/INTEGE/IFTA,BETA,GAMA,RTOL,NGAMA,NPUNCH,NEQUIB,ITEMX
COMMON/PETER/AREA,WIDTH,HOK,VERT,ACCG
COMMON/ASIM/NYP,NSC,NSLC,NLC,NNYP,NNSC,ISLC

DIMENSION X(1),AU2(1)
*****DYNAMIC LOAD MATRIX FORMULATION *****

DYNAMIC LOADS

IF(ISLC.EQ.0) GO TO 900
IF(NC.NE.0) GO TO 700

45 CONTINUE
  DO 50 N=1,2
  DO 50 I=1,NUMNP
  DO 50 J=1,4
50 PDYL(N,J,1)=0.0
  READ(5,1005) HED
  READ(5,1006) NUMLP
  WRITE(6,2005) HED,NUMLP
  IF(NUMLP.EQ.0) GO TO 500
  KL=1
55 DO 60 M=KL,2
  READ(5,1012) TD(N)
  WRITE(6,2016) TD(N)
  DO 60 I=1,NUMLP
  READ(5,1007) M,(PDYL(N,J,M),J=1,4)
60 WRITE(6,2007) M,(PDYL(N,J,M),J=1,4)
  IF(IPLNAX.NE.1) GO TO 80
  DO 70 N=1,2
  DO 70 I=1,NUMNP
  DO 70 J=1,4
  XX=X(I)
70 PDYL(N,J,1)=PDYL(N,J,1)+XX
80 CONTINUE
  IF(KL.EQ.2) GO TO 117
  KL=2
  I=2
  DTXX=TD(2)-TD(1)
  TM=TD(1)+DT

SET UP LOAD MATRIX

```

```

00001**2
00002**5
00003
00004**5
00005**5
00006**5
00007
00008
00009
00010**9
00011**9
00012**9
00013**9
00014**8
00015**5
00016**5
00017**3
00018
00019
00020
00021
00022**8
00023**8
00024**8
00025
00026
00027
00028
00029
00030
00031
00032
00033
00034
00035
00036
00037
00038
00039
00040
00041
00042
00043
00044
00045
00046
00047**2
00048**2
00049
00050
00051
00052
00053
00054
00055
00056
00057
00058
00059

```



```

00060
00061
*****CREATE ACTUAL LOAD VECTOR FOR DISCRETE TIME ST00062
FOR DYNAMIC LOAD 00063
700 CONTINUE 00064
IF (NUMNP.EQ.0) GO TO 900 00065
IF (TM.GT.TK) GO TO 120 00066
IF (TD(2)-TM) 115,120,120 00067
115 TD(1)=TD(2) 00068
DO 116 N=1,4 00069
DU 116 M=1,NUMNP 00070
116 PDYL(1,N,M)=PDYL(2,N,M) 00071
GO TO 55 00072
117 DTXX=TD(2)-TD(1) 00073
IF (DTXX) 118,115,120 00074
118 WRITE(6,2010) TD(2) 00075
GO TO 900 00076
120 DO 123 I=1,NUMNP 00077
DU 123 J=1,4 00078
JJ =ID(1,J) 00079
IF (TM.GT.TK) GO TO 129 00080
IF (JJ) 123,123,122 00081
122 SLOPE=(PDYL(2,J,1)-PDYL(1,J,1))/DTXX 00082
AU2(JJ)=PDYL(1,J,1)+(TM-TD(1))*SLOPE 00083
GO TO 123 00084
129 CONTINUE 00085
IF (JJ) 123,123,128 00086
128 AU2(JJ)=PDYL(2,J,1) 00087
123 CONTINUE 00088
130 CONTINUE 00089
WRITE(6,190) 00090
190 FORMAT(1H1/' *** LOAD VECTOR IN LOAD2 ***** ') 00091
WRITE(6,195) NC,(AU2(I),I=1,NEQ) 00092
195 FORMAT(1X,12,16E8.1) 00093**6
900 CONTINUE 00094**7
RETURN 00095**6
1005 FORMAT (10A8) 00096**6
1006 FORMAT (15) 00097
1007 FORMAT(15,4F10.0) 00098
1012 FORMAT(8F10.0) 00099
2005 FORMAT (1H1,/,8A8,/, 00100
* ' NUMBER OF NODES SUBJECT TO DYNAMIC LOAD = ',15) 00101
2007 FORMAT (15,4F10.4) 00102
2010 FORMAT(/'ERROR IN BASE ACCEL OR DYN LOAD,TIME = ',F10.3) 00103
2016 FORMAT (1H1/' NODAL POINT LOADS AT TIME = ',F10.3//) 00104
END 00105
00106
00107
00108
00109
00110

```

```

DATA SET LOAD3      AT LEVEL 009 AS OF 83/02/11
SUBROUTINE LOAD3(IS,GACL,DI,NACL,XM,LD,NUMNP,NC,TK,TM,AU3,ACCL) 00001**2
IMPLICIT REAL*8 (A-H,O-Z) 00002**4
DIMENSION TS(1), GACL(1),LD(2),AU(1),LD(NUMNP,1),XM(1) 00003
COMMON/CHNTL/DD,HED(8),RADN,NASIN,NUMSH,NKK,MBAND,NCYCL,NBYN,IPLNA 00004**4
*X,ISTRPR,NPKINP,ISTART,NEQK,MBANDH 00005
*,NEQS,MBANDS 00006
COMMON/PETER/AREA,WIDTH,HOR,VERT,ACCG 00007
DIMENSION AU3(1) 00008
IF(NC.NE.0) GO TO 700 00009**9
READ BASE ACCEL. HISTORY 00010**9
READ(5,1004) HED 00011
READ(5,1009) FAC1,FAC2 00012**2
READ(5,1008) (TS(1),GACL(1),I=1,NACL) 00013**4
TS(1)=0.0 00014**4
GACL(1)=0.0 00015
NIN=NACL-1 00016
IN=2 00017
READ(5,1008) (GACL(1),I=1N,NACL) 00018
MIN=1 00019
DO 251 L=2,NACL 00020
TS(L)=TS(L-MIN)+DI 00021
WRITE(6,2008) HED ,FAC1,FAC2 00022
DO 40 I=1,NACL 00023
TS(I)=TS(I)*FAC1 00024**8
40 GACL(I)=GACL(I)*FAC2 00025**8
WRITE(6,2009) ((TS(N),GACL(N)),N=1,NACL) 00026**8
*****END ***** 00027**8
100 CONTINUE 00028
TM=TS(1)+DI 00029
L=2 00030
DTX=TD(L)-TD(L-1) 00031
PRINT 999 00032
WRITE(6,2170) ((A3(I,J),J=1,MBAND),I=1,NEQ) 00033
PRINT 999 00034
WRITE(6,2170) (A4(I ),I=1,NWA) 00035
PRINT 999 00036
WRITE(6,2170) ((A5(I,J),J=1,MBAND),I=1,NEQ) 00037
2170 FORMAT(5X,10E12,5) 00038
*****CREATE LOAD VECTOR FOR GROUND MOTION DISCRETE STEP 00039
00040
00041
00042
00043
00044
00045
00046
00047
00048
00049
00050
00051
00052
00053
00054
00055
00056
00057
00058
00059

```

```

FOR BASE MOTION                                00060
700 CONTINUE                                  00061
150 IF (TS(L)-TM) 155,155,160                 00062
155 CONTINUE                                  00063
    IF(L.EQ.NACL) GO TO 161                   00064
    DTX=TS(L+1)-TS(L)                          00065
    L=L+1                                       00066
    IF (DTX) 158,155,160                       00067
158 WRITE(6,2010) TS(L-1)                     00068
    GO TO 900                                   00069
160 SLOPE=(GACL(L)-GACL(L-1))/DTX              00070
    ACCL=GACL(L-1)+(TM-TS(L-1))*SLOPE          00071
    GO TO 162                                   00072
161 ACCL=GACL(L)                               00073
162 DO 180 I=1,NOMNT                           00074
    DO 170 J=1,2                                00075
    JJ=10(I,J)                                  00076
    IF (JJ) 170,170,165                        00077
165 AU3(JJ)=0.0                                 00078
    IF (JJ.EQ.10(1,1)) AU3(JJ)=-ACCL*XM(JJ)   00079
170 CONTINUE                                  00080
180 CONTINUE                                  00081
                                           00082
    ACCG=ACCL                                  00083
    WRITE(6,190)                                00084**9
190 FORMAT(1H1/'      *** LOAD VECTOR IN LOAD3 ***** ') 00085**6
    WRITE(6,195) NC,(AU3(I),I=1,NEG)           00086**7
195 FORMAT(1X,12,16E8.1)                       00087**6
                                           00088**6
    00089
900 CONTINUE                                  00090
    RETURN                                       00091
1004 FORMAT (10A8)                              00092
1008 FORMAT(8F9.6)                              00093
1009 FORMAT(2F8.3)                              00094
2008 FORMAT(1H1/10A8/                          00095**3
* ' SCALE FACTOR FOR TIME                      = 'F10.3/
* ' SCALE FACTOR FOR ACCEL                     = 'F10.3//)
2009 FORMAT (/5X,'POINT',11X,'TIME',11X,'ACCEL',/(5X,15,2F15.6)) 00098
2009 FORMAT(1H1/6(1X,11HLINE VALUE,2X,8HFUNCTION),/(6(F5.2,E12.6))) 00099**5
2010 FORMAT(/'ERROR IN BASE ACCEL OR DYN LOAD,TIME = 'F10.3) 00100
    END                                         00101

```

DATA SET LDDCHK	AT LEVEL 002 AS OF 82/07/14	00001
SUBROUTINE LDDCHK(NDT,NUMEL,NLC,NPJ,NEQS,CT,SN,SU)		00002
IMPLICIT REAL*8 (A-H,O-Z)		00003
COMMON/ASIM/NYP,NSC,NSIC,NGC,NNYP,NNSC		00004
DIMENSION RM(5),ZM(5),SIG(4),STRAIN(4),SU(1),CT(NPJ,1),SN(NPJ)		00005
		00006
		00007**2
		00008
		00009
		00010
REWIND 55		00011
DO 300 LC=1,NLC		00012
		00013
DO 100 M=1,NUMEL		00014
		00015
DO 200 I=1,NDI		00016
		00017
		00018
		00019
READ(55) L,RH(L),ZM(L),(SIG(I),I=1,4),(STRAIN(JJ),JJ=1,4),EPSV,		00020
*EPSVF		00021
		00022
K1=(NSC+I-2)*(M-1)+NSC+NDI		00023
SN(K1)=SIG(1)		00024
SN(K1+1)=SIG(2)		00025
SN(K1+2)=SIG(4)		00026
		00027
200 CONTINUE		00028
		00029
100 CONTINUE		00030
		00031
CALL CHECK(CT,SN,SU,NPJ,NEQS)		00032
		00033
300 CONTINUE		00034
RETURN		00035
END		00036

DATA SET MAIN AT LEVEL 103 AS OF 83/02/28
 PROGRAM FSDYN (INPUT,OUTPUT,TAPE3,TAPE4,TAPE5=INPUT,TAPE6=OUTPUT) 00001
 SUBROUTINE MAIN(MCOM)

IMPLICIT REAL*8 (A-H,O-Z)	00002
COMMON D(400000)	00003
	00004*81
	00005102
	00006*81
	00007*17
	00008*17
	00009*17
COMMON/EMB/DBDH(1500)	00010*17
COMMON/QUAD/CCCC(1500),N1,N2,N3,N4,N5,MTOT,NNN,NK15	00011*68
COMMON/CONTL/DT,HFD(8),KADN,NUMNP,NUMEL,NEG,MBAND,NCYCL,NDYN,IPLNA	00012
*X,ISTRPR,NPRINT,ISTART,NEQH,MBANDH,NEQS,MBANDS	00013**8
COMMON/INTEGR/ETA,BETA,GAMA,RTOL,NGAMA,NPUNCH,NEQUIB,ITEMX	00014*78
COMMON/SOL/ANORM,NBLUCK,NEQB,LL,NF,IDUM,NEIG,NAD,NVV,NFO	00015
COMMON/EXTRA/MODEX,NTB,N1USV,NT10	00016
COMMON/TAPES/NOQ(6),NUMBER,LCOUNT	00017101
COMMON/DIMED/N43,IP1,IP11,KK,NPKN1	00018*90
COMMON/BFLA/NS,NELTYP,ND,NUMGT,MODEL,NCUN,IDW,NK015,NK016,NK017	00019*81
	00020*44
	00021
COMMON/DENBY/DB(5,200,4,4),NCOND,NTENS	00022*99
	00023*98
DIMENSION SIG(9)	00024*98
COMMON/GORA/PLAG	00025*98
	00026*98
	00027
	00028
	00029*77
COMMON/BUCK/TIM1,TIM2,TIM3	00030*77
	00031*67
	00032*67
COMMON/BOBBY/KKS,NUMMAT,NC,LOUP,NDRN,NINT,IDWA,NPT	00033*67
COMMON/ANON/TA	00034*22
	00035*46
***** ALL INFORMATION ABOUT PLOTTING *****	00036*46
	00037*46
COMMON/KABI/JPT,KEL,JPLDT(200),KELMT(200)	00038*94
COMMON/PETER/AREA,WIDTH,HOR,VERT,ACCG	00039101
COMMON/NEVEL/WEIGHT,SHEAR	00040101
	00041*94
	00042*94
	00043*77
***** FINISH ALL INFORMATION REGARDING PLOT *****	00044*77
	00045*77
	00046*77
	00047*46
	00048*46
DIMENSION IDUMM(36)	00049
EQUIVALENCE (IDUMM(1),D(1))	00050
	00051*43
	00052*43
COMMON/ASIM/NYP,NSC,NSLC,NLC,NNYP,NNSC	00053
	00054
	00055*77
CALL SETIME	00056*80
	00057*77
MTOT=400000	00058103

		00059*94
		00060*94
885	DO 885 I=1,NIUT	00061*86
	D(I)=0.0	00062*86
	P1=4.0*DAIAN(1.000)	00063
	RADN=180./P1	00064
	LOOP=1	
		00065
5	CONTINUE	00066
	CALL ERKSH(88,.TRUE.,.FALSE.,.FALSE.,.FALSE.,255)	00067
	CALL ERKSH(89,.TRUE.,.FALSE.,.FALSE.,.FALSE.,255)	00067
	CALL ERKSH(208,256,-1,1,1,208)	00068
	READ CONTROL DATA	00069
		00070
		00071
		00072*77
	CALL GETIME(TIMI)	00073*77
	TIMI=FI,0AT(1TIMI)/1000.	00074*80
		00075*80
		00076*81
	WRITE(6,1008) TIMI	00077*81
	WRITE(6,1003) TIMI	00078*81
		00079*77
		00080*77
	READ(5,990) HED	00081*55
	READ(5,1000) NUMNP,NUMEL,NDYN,IPLNAX,LL,NF,MODEX,NAD,NGAMA	00082*73
	IF(MODEX.GT.0) MODEX=1	00083
	IF(NUMNP.EQ.0) STOP	00084
	WRITE(6,2000) HED,NUMNP,NUMEL,NDYN,IPLNAX,LL,NF,MODEX,NAD	00085*74
	READ(5,1002) ISTRPR,NPRINT,ISTART	00086
	READ(5,1001) NCYCL,DT,KKS,FA,NDRN,NCUN,IDW,MODEL,NINT	00087*67
	WRITE(6,2001) NCYCL,DT,KKS,FA,NDRN,NGAMA	00088*74
		00089
		00090*45
		00091*45
	READ(5,1020) AREA,WIDTH,WEIGHT,SHEAR	00092*101
	WRITE(6,1030) AREA,WIDTH,WEIGHT,SHEAR	00093*101
1030	FORMAT(10I1//40X,' PLATFORM AREA = ',E10.3/	00094*101
*	40X,' PLATFORM WIDTH = ',E10.3/	00095*101
*	40X,' PLATFORM WEIGHT = ',E10.3/	00096*101
*	40X,' SOIL SHEAR STRENGTH= ',E10.3/)	00097*101
1020	FORMAT(4E10.3)	00098*101
	***** INPUT AND DATA RELATED PLOTTING OF OUTPUT *****	00099*45
	READ(5,1008) JPI	00101*94
	IF(JPI.EQ.0) GO TO 550	00102*94
	READ(5,1009) (JPLUT(I),I=1,JPI)	00103*94
550	READ(5,1011) KEL	00104*94
	IF(KEL.EQ.0) GO TO 560	00105*94
	READ(5,1009) (KELMI(I),I=1,KEL)	00106*94
		00107*101
		00108*101
1050	WRITE(6,1050)	00109*101
	FORMAT(10I1// ' ***** NODE AND ELEMENT DATA *** '/)	00110*101
	WRITE(6,1040) JPI,(JPLUT(I),I=1,JPI)	00111*101
	WRITE(6,1040) KEL,(KELMI(I),I=1,KEL)	00111*101
1040	FORMAT(10X,20I5)	00112*101
560	CONTINUE	00113*94
		00114*94
		00115*94
		00116*94

```

***** FINISHED READING ALL INPUT DATA RELATED TO PLOTTING
CONTINUE
INPUT NODAL POINTS
N1=1
N2=4*NUMNP+N1
N3=NUMNP+N2
N4=NUMNP+N3
IF (N4.LI.NTOT) GO TO 810
WRITE(6,2006)
WRITE(6,815)
815 FORMAT(1H1/' *** VALDE OF N4 **** ')
WRITE(6,820) N4
STOP
820 FORMAT(2X,15)
810 CONTINUE

*****
SUBROUTINE READS INPUT DATA AND FINDS THE TOTAL EQUAT
*****
***** N1      =    ID
***** N2      =    X
***** N3      =    Y
***** N4      =    IH
CALL INPUTJ(D(N1),D(N2),D(N3),D(N4),NUMNP,NEG,NEGH,NEGS)
N5=N4+NUMNP*2

CALL GETIME(T1,T2)
TIM2=FLOAT(T1-T2)/1000.

WRITE(6,1008) T1-T2
WRITE(6,1003) TIM2

*** PROCESSING OF BASIC INPUT DATA  NODAL POINTS  ***
PINPUT=(TIM2-T1)

*****
SUBROUTINE ELEMENT CALCS LEFT AND PASSES DIFFERENT STIFFMA
MASSES, AND DAMPING ,LOAD VECTOR FOR STATIC OR DYNAMIC ANAL
AND WRITES THOSE ON TAPES

```

```

00117*46
00118*46
00119*46
00120*45
00121*45
00122*45
00123
00124
00125
00126
00127
00128
00129
00130*69
00131*69
00132*69
00133*69
00134*69
00135*69
00136*69
00137*69
00138*69
00139*69
00140
00141
00142
00143
00144
00145
00146**8
00147**8
00148*28
00149*28
00150*28
00151*28
00152*28
00153**8
00154
00155
00156
00157
00158*77
00159*80
00160*80
00161*81
00162*81
00163*81
00164*81
00165*77
00166*77
00167*77
00168*77
00169*77
00170
00171
00172
00173
00174
00175
00176

```

```

*****
00177
00178
00179*17
00180*17
00181*98
00182*98
00183*17
00184
00185*35
00186*35
00187*35
00188*35
00189*35
00190*35
00191*35
00192*35
00193*17
00194
00195*69
00196*77
00197*77
00198*80
00199*80
00200*77
00201*81
00202*81
00203*77
00204*77
00205*77
00206*77
00207*77
00208*69
00209*69
00210*69
00211*70
00212*70
00213*69
00214*69
00215*69
00216*69
00217*94
00218*94
00219
00220*33
00221*33
00222101
00223*33
00224*33
00225*35
00226
00227
00228*11
00229*11
00230
00231
00232
***** ALLOCATE RESIDUAL LOAD VECTOR AT THIS POINT *****00233*45
***** NPO12 IS THE STARTING POINT FOR RESIDUAL LOAD VECT00234*45
NPO12=N011+(NEQ+MBAND-1)
NPO12=NPO12+NEQ
00235*73
00236*73

```

```
IFLAG=0
```

```
NNN=1
NC=0
```

```
***** TAPE 14 REQUIRED FOR ELASTO PLASTIC ANALYSIS
```

```
REWIND 14
```

```
CALL ELEM1(NUMMAT)
```

```
CALL GETIME(ITIM3)
TIM3=FLOAT(ITIM3)/1000.
```

```
WRITE(6,1008) ITIM3
WRITE(6,1003) TIM3
```

```
*** PROCESSING OF BASIC ELEMENT DATA INITIALIZATION
*** ELEMENT STIFFNESS, MASS, LOAD VECTOR FORMULATION
```

```
PELEMT=(TIM3-TIM2)
```

```
IF(NNN.EQ.1) GO TO 830
WRITE(6,2006)
WRITE(6,840)
```

```
840 FORMAT('HI!', ' *** PROBLEM IN ELEM1 SUBROUTINE! ')
WRITE(6,820) NK15
STOP
```

```
830 CONTINUE
READ(5,1003) TETA,BETA,GAMA,RTOL,NPUNCH,NEQUIB,ITEMX,NACL
WRITE(6,2008)TETA,BETA,GAMA,RTOL,NPUNCH,NEQUIB,ITEMX
IF(NACL.NE.0) WRITE(6,2002) NACL
```

```
READ(5,505) NLC,NP3,NP6,NUMBER,NCOND,NTENS
```

```
N6=NEQS+MBANDS+NK15
N7=NEQH+MBANDH+NO
N07=N7+NEQ
N8=N07+NEG
N09=N8+NEQ*MBAND
N010=N09+NEQ*MBAND
N011=N010+NEQ*MBAND
```

```
***** ALLOCATE RESIDUAL LOAD VECTOR AT THIS POINT *****
***** NPO12 IS THE STARTING POINT FOR RESIDUAL LOAD VECT
NPO12=N011+(NEQ+MBAND-1)
NPO12=NPO12+NEQ
```



```

NO12=NO012+(NEG)                                00237*73
                                                    00238*69
                                                    00239*69
                                                    00240*69
IF(NO12.L1.M101) GO TO 860                        00241*70
WRITE(6,2006)                                     00242*69
WRITE(6,850)                                       00243*69
850 FORMAT(1H1/1' **** VALUE OF NO12 ****')      00244*70
      WRITE(6,820) NO12                            00245*69
STOP                                               00246*69
860 CONTINUE                                       00247*70
                                                    00248
                                                    00249*17
                                                    00250*17
                                                    00251*17
NN07=NO12+NUMMAT                                  00252*17
NN08=NN07+NUMMAT                                  00253*17
NN09=NN08+NUMMAT                                  00254*17
NN10=NN09+NUMMAT                                  00255*17
NN11=NN10+NUMMAT                                  00256*17
NN12=NN11+NUMMAT                                  00257*17
NN13=NN12+NUMMAT                                  00258*17
NN14=NN13+NUMMAT                                  00259*17
NN15=NN14+NUMMAT                                  00260*33
                                                    00261*17
ND15=NN15+NEQS*NEQS
IF(KKS.EQ.2) GO TO 835
ND16=ND15+NEQS*1
ND17=ND16+NEQS
ND18=ND17+NEQS*1
ND19=ND18+NEQS
GO TO 836
                                                    00262*17
835 CONTINUE
ND16=ND15+NEQS*NEQS
ND17=ND16+NEQS
ND18=ND17+NEQS*NEQS
ND19=ND18+NEQS
836 CONTINUE
                                                    00263*17
***** NN07 = GAX                                00264*28
***** NN08 = DPL                                00265*28
***** NN09 = GUSE                                00266*28
***** NN10 = E                                   00267*28
***** NN11 = EN                                  00268*28
***** NN12 = PU                                  00269*28
IF(KKS.EQ.2) CALL STRCOM(NUMMAT,D(NN07),D(NN08),D(NN09),D(NN10),
00270*17
00271*41
00272*41
*D(NN11),D(NN12))                                00273*20
00274*78
00275*78
CALL GETIME(I1IM4)                                00276*80
TIM4=FLOAT(I1IM4)/1000.                          00277*80
00278*78
00279*78
SUM=TIM4-TIM3                                     00280*78

```

```

NDUM=NUMM1
NUMBER=1
SUM1=0.
SUM2=0.
SUM3=0.
SUM4=0.
SUM5=0.
DO 600 LOOP=1,NUMBER
LCOUNT=0

CALL GETIME(TIM5)
TIM5=FLOAT(TIM5)/1000.

IF (KKS.EQ.2.OR.KKS.EQ.4)
* CALL SUB(D(NK15),D(N07),NEQB,MBANDS,NUMNP,NUMEL,
*KKS,D(N1),D(N4),LOOP,NDUM,D(NN13),D(NN14),NUMMAT,D(NN07)
*,D(NN08),D(NN09),D(NN10),D(NN11),D(NN12),D(NN15),D(ND15),D(ND16),
*D(ND17),D(ND18))

CALL GETIME(TIM6)
TIM6=FLOAT(TIM6)/1000.

SUM1=SUM1+(TIM6-TIM5)
*****
SUBROUTINE KSTAR ASSEMBLE THE DIFFERENT MATRICES FORMED
ELSF
*****
CALLKSTAR(D(N4),D(NK15),D(N6),D(N7),NEQ,MBAND,NEQB,MBANDH,D(N8),
*D(N09),D(N010),D(N011),NWA,NEQB,NBLOCK,MI,D(N1),D(NP012),D(ND012))

CALL GETIME(TIM7)
TIM7=FLOAT(TIM7)/1000.

*** PROCESSING AND ASSEMBLING DECOMPOSING STIFFNESS MATRIX ***
SUM2=SUM2+(TIM7-TIM6)

```

```

00281*78
00282*31
00283*33
00284*77
00285*77
00286*77
00287*77
00288*77
00289*31
00290*78
00291*31
00292*31
00293*31

```

```

00294*31
00295*17
00296*80
00297*80
00298*78
00299*17
00300*17
00301**9
00302*37
00303*41
00304*33
00305*41
00306*20
00307**8

```

```

00308**8
00309
00310*78
00311*80
00312*80
00313*78
00314*78
00315*78
00316

```

```

00317
00318
00319
00320
00321
00322
00323
00324
00325*38
00326*41
00327*73
00328

```

```

00329*77
00330*77
00331*80
00332*80
00333*78
00334*77
00335*77
00336*77
00337*78

```

```

                                00338*78
                                00339*77
                                00340*77
                                00341
                                00342
NN17=ND19+1                    00343*33
IF(LOOP.NE.1) GO TO 650        00344*31
                                00345*69
                                00346*69
                                00347*69
                                00348*33
701 N9=NEQ+NN17                00349
N10=NEQ+N9                      00350
N11=NEQ+N10                      00351
N12=2*4+NUMNP+N11                00352
IF(NDYN.NE.1) N12=N11            00353
N13=NACB+N12                      00354
N14=NACL+N13                      00355
N15=NEQ+N14                      00356
N16=NEQ+N15                      00357
N17=NEQ+N16                      00358
NO17=N17+NEQ                     00359*17
N18=18*NUMEL+NO17                00360*97
N19=24*NUMNP+N18                 00361*33
                                00362*33
WRITE(6,508) N1,N2,N3,N4,N5,N6,N7,N07,N8,N09,N010,N011,N012,
*NN07,NN08,NN09,NN10,NN11,NN12,NN13,NN14,NN15,NN17,N10,N11,N12,N13,
                                00363*41
                                00364*41
                                00365*33
                                00366*41
                                00367*41
                                00368*33
* N14,N15,N16,N17,N18,N19        00369*43
508 FORMAT(1H1/ ' N1 TO N19 ',/ ,2(1710,/))
                                00370*41
                                00371*41
                                00372
N20 =2*NEQ+N19                    00373
N25=N20+NEQ                      00374
N26=N25+NEQ                      00375
N27=N26+NEQ                      00376
N28=N27+NEQ                      00377
N29=N28+NEQ                      00378
N30=N29+NEQ                      00379
NO29=N29+NEQ                     00380
N30=NO29+NEQ                     00381*69
                                00382*69
                                00383*72
IF(N30.LT.N101) GO TO 875        00384*69
WRITE(6,2006)                    00385*69
WRITE(6,870)                      00386*69
870 FORMAT(1H1/'***** VALUE OF NI -*** ')
WRITE(6,820) N30                  00387*69
STOP                               00388*69
                                00389*18
                                00390*18
                                00391*18
                                00392*18
5000 II=ND                         00393
NASIM=N21-1                       00394
DO 1500 I=N20,NASIM              00395
D(I)=D(II)                       00396
1500 II=II+1                      00397

```

```

      II=N09
      NAKIM=N22-1
      DO 1501 I=N21,NAKIM
1501  D(I)=D(II)
      II=II+1
      II=N010
      NADIM=N23-1
      DO 1502 I=N22,NADIM
1502  D(I)=D(II)
      II=II+1
      II=N011
      NAFIM=N24-1
      DO 1503 I=N23,NAFIM
1503  D(I)=D(II)
      II=II+1

875  CONTINUE
      N21=0
      N22=0
      N23=0
      N24=0

*****
      SUBROUTINE SOLVED CALLS STRESS AND FINDS THE SOLUTIONS
*****
625  CONTINUE
      WRITE(6,509)  NLC,NP3,NP6,NUMBER,NCOND,NTENS
      NIG=NLC-1
      NDT=1
      IF(KKS.EQ.4.AND.NCOND.EQ.1) NDT=4

      NXP=NP6
      NSC=NP3
*****  VARIABLE DEFINED FOR ALLOCATION *****
      NPJ=NPJ+NUMEL
      NPK=NP6*NUMEL

      NPJ=NPJ+NDT
      NPK=NPK+NDT

      N31=N30+NUMNP*4
      N32=N31+NUMNP
      N33=N32+NUMNP

*****  TEMPORARY ALLOCATION FOR SURFACE LOAD VECTOR
      N34=N33+2*LL*2
      N35=N34+2*LL*2

```

```

00398
00399
00400
00401
00402
00403
00404
00405
00406
00407
00408
00409
00410
00411
00412
00413
00414
00415
00416
00417
00418100
00419
00420103
00421103
00422103
00423103
00424
00425
00426
00427
00428
00429
00430*33
00431*33
00432*33
00433*57
00434*33
00435
00436
00437
00438
00439
00440
00441*82
00442*82
00443*82
00444*82
00445*82
00446
00447
00448
00449*59
00450*59
00451*59
00452*59
00453*59
00454*73
00455*73

```

```

NNYP=NUMEL*NYYP                                00456
NNYP=NNYP*ND1                                    00457*82
N36=N35+NNYP                                    00458*82
N37=N36+NNYP                                    00459*82
NNSC=NUMEL+N3C                                  00460
NNSC=NNSC*ND1                                    00461
N38=N37+NNSC*NNYP                                00462
N39=N38+ND1C                                     00463*82
N40=N39+NYYP*ND1                                 00464*82
N41=N40+NYYP                                      00465*82
***** TEMPORARY ALLOCATION FOR SURFACE PORE PRESSURE 00466*41
N42=N41+Z*LL*Z                                    00467
N43=N42+Z*LL*Z                                    00468*61
IF(N43.LT.MID1) GO TO 880                         00469
WRITE(6,2006)                                      00470*59
WRITE(6,890)                                       00471*59
890 FORMAT(1H1/' *** VALUE OF N43 *** ')          00472*59
WRITE(6,820) N43                                    00473*74
STOP                                               00474*74
880 CONTINUE                                       00475*59
***** CREATE HERE DO LOOP FOR STATIC SHAKEDOWN P00486
IF(KKS.EQ.4) READ(5,500) IPT,IPT1,KK,NPRNT       00476*69
650 CONTINUE                                       00477*69
***** ASSIGN ALL TAPE NUMBERS FOR OUTPUT ***** 00478*70
TAPE 71 USED IN 'ASSEMB' AND 'ITRSLD' ROUTINE    00479*69
TAPE 55 USED IN 'OUT' ROUTINE TO STORE DISPL.,VEL.,ACCLN. 00480*69
TAPE 57 USED IN 'STRDR' ROUTINE TO STORE STRESSES ,STRAINS 00481*69
REWIND 17                                          00482*69
REWIND 18                                          00483*69
REWIND 21                                          00484*69
REWIND 55                                          00485
REWIND 57                                          00486
REWIND 61                                          00487

```

DO 8000 IC=1,NIC

CALL GETTIME(TIME)

TIME=FLOAT(TIME)/1000.

WRITE(6,7490) N20,N21,N22,N23,N24,N25,N26,N27,N28,N29,N30,N31,
 *N32,N33,N34,N35,N36,N37,N38,N39,N40
 7490 FORMAT(1H1/1 N20 10 N40 1,/,2(10I12,/))

*****	N007	=	B	00516
*****	N9	=	BO	00517*78
*****	N10	=	VEL	00518*80
*****	N11	=	PHI	00519*80
*****	N12=	=	TS	00520*78
*****	N13	=	GACL	00521*33
*****	N1	=	ID	00522*33
*****	N14	=	K	00523
*****	N15	=	A	00524
*****	N16	=	AO	00525*86
*****	N17	=	MB	00526*18
*****	N017	=	EMAX	00527*18
*****	N18	=	AMAX	00528*18
*****	N19	=	MB3	00529*18
*****	N8	=	AB	00530*28
*****	N09	=	AB	00531*28
*****	N010	=	AO	00532*28
*****	N011	=	MAXA	00533*28
*****	N20	=	AO	00534*28
*****	N25	=	A1	00535*28
*****	N26	=	A2	00536*28
*****	N27	=	AT	00537*28
*****	N28	=	BI	00538*28
*****	N29	=	AC	00539*28
*****	N2	=	X	00540*28
*****	N3	=	Y	00541*28
*****	N30	=	Q	00542*28
*****	N31	=	JSC	00543*28
*****	N31	=	JSC	00544*28
*****	N33	=	SURTRX	00545*28
*****	N34	=	SURTRY	00546*28
*****	N35	=	AN	00547*28
*****	N36	=	EK	00548*28
*****	N37	=	AN	00549*28
*****	N38	=	NSLC1	00550*28
*****	N39	=	ANI	00551*28
*****	N40	=	EKJ	00552*28
*****	N41	=	SURPF X	00553*28
*****	N42	=	SURPF Y	00554*28
*****	N012	=	RE	00555*28
*****	N012	=	SW	00556*29
*****				00557*29
*****				00558*28
*****				00559*28
*****				00560*28
*****				00561*28
*****				00562*28
*****				00563*28
*****				00564*28
*****				00565*28
*****				00566*28
*****				00567*28
*****				00568*28
*****				00569*28
*****				00570*28
*****				00571*33
*****				00572*76
*****				00573*76
*****				00574*33

```

WRITE(6,8001) N4,NK15,N6,N7,N007,N9,N10,N11,N12,N13,N1,N14,N15,N16
WRITE(6,8002)
*N17,N017,N18,NEQ,NUMNP,NACL,NUMEL,NEQH,N19,N8,N09,N010,N011,N20
WRITE(6,8003)
*N25,N26,N27,N28,N29,N029,NWA,NEQB,NBLUCK,M1,N2,N3,N30,N31,N32,N33
WRITE(6,8004)
*N34,N35,N36,N37,N38,N39,N40,LC,N41,N42,NP012,ND012
    
```

00575

```

8001 FORMAT(1H1/
* LOCATION OF I1(NUMNP,1) N4 = ,18/
* LOCATION OF A1(NEGS,1) NK15 = ,18/
* LOCATION OF A3(NEG,1) N6 = ,18/
* LOCATION OF XM(NEQ) N7 = ,18/
* LOCATION OF B(NEQ) N007 = ,18/
* LOCATION OF BU(NEQ) N9 = ,18/
* LOCATION OF VED(NEG) N10 = ,18/
* LOCATION OF PDYL(2,4,NUMNP) N11 = ,18/
* LOCATION OF TS(NACL) N12 = ,18/
* LOCATION OF GACL(NACL) N13 = ,18/
* LOCATION OF ID(NUMNP,1) N1 = ,18/
* LOCATION OF R(NEQ) N14 = ,18/
* LOCATION OF A(NEQ) N15 = ,18/
* LOCATION OF AU(NEQ) N16 = ,18/
8002 FORMAT(
* LOCATION OF GB(NEQ) N17 = ,18/
* LOCATION OF PMAX(NUMEL,1) = ,18/
* LOCATION OF AMAX(NUMNP,12,2) = ,18/
* NEQ = ,18/
* NUMNP = ,18/
* NACL = ,18/
* NUMEL = ,18/
* NEQH = ,18/
* LOCATION OF GCB(NEQ) N19 = ,18/
* LOCATION OF A4(NEQ,1) N8 = ,18/
* LOCATION OF A5(NEQ,1) N09 = ,18/
* LOCATION OF A6(NWA) N010 = ,18/
* LOCATION OF MAXA(NEQ) N011 = ,18/
8003 FORMAT(
* LOCATION OF X0(NEQ) N20 = ,18/
* LOCATION OF X1(NEQ) N25 = ,18/
* LOCATION OF X2(NEQ) N26 = ,18/
* LOCATION OF A1(NEQ) N27 = ,18/
* LOCATION OF R1(NEQ) N28 = ,18/
* LOCATION OF AC(NEQ) N29 = ,18/
* LOCATION OF S0(NEQ) N029 = ,18/
* NWA = ,18/
* NEQB = ,18/
* NBLUCK = ,18/
* M1 = ,18/
* LOCATION OF X(NUMNP) N2 = ,18/
* LOCATION OF Y(NUMNP) N3 = ,18/
* LOCATION OF Q( ) N30 = ,18/
* LOCATION OF ISC(NUMNP) N31 = ,18/
* LOCATION OF JSC(NUMNP) N32 = ,18/
8004 FORMAT(
* LOCATION OF SURTRX(2,LL,2) N33 = ,18/
* LOCATION OF SURTRY(2,LL,2) N34 = ,18/
* LOCATION OF AM(NNYP) N35 = ,18/
* LOCATION OF ER(NNYP) N36 = ,18/
* LOCATION OF AN(NWSC+NNYP) N37 = ,18/
* LOCATION OF NSLC1(NLC) N38 = ,18/
* LOCATION OF AN(LLNYP*NNYP) N39 = ,18/
* LOCATION OF ER0(NYP) N40 = ,18/
    
```

```

*          :          LC          = ',18/
*          : LOCATION OF SURFFX(2,LL,2)N41 = ',18/
*          : LOCATION OF SURFFY(2,LL,2)N42 = ',18/
*          : LOCATION OF RE(NEQ) NPO12= ',18/
*          : LOCATION OF SW(NEQ) ND012= ',18/)
CALL SOLVED(D(N4),D(NK15),D(N6),D(N7),D(N007),D(N9),D(N10), 00576*42
00577*43
*D(N11),D(N12),D(N13), D(N1),D(N14), D(N15),D(N16),D(N17),D(N0100578
00579
*7),D(N18), NEQ,NUMNF,NACL,NUMEL,NEQH,D(N19),D(N8),D(N09),D(N00580
00581
*D10),D(N011),D(N20),D(N25),D(N26),D(N27),D(N28),D(N29),D(N029), 00582
00583
* NWA,NEQB,NBLOCK,MI, D(N2), 00584
00585
*D(N3), D(N30),D(N31),D(N32),D(N33),D(N34),D(N35),D(N36),D(N3700586
00587*33
00588*33
*),D(N38),D(N39),D(N40),LC,D(N41),D(N42),D(NPO12),D(ND012),D(NN15))00589*73
00590*26
00591*76
00592*76
00593*76
***** PUT CONDITION FOR GRAVITY STATIC ANALYSIS WITH DYNAM00594*53
SHAKEDOWN PROBLEM ***** CALL KSTAR AND RESET 00595*53
00596*53
GO TO (755,755,755,758),KKS 00597*56
758 CONTINUE 00598*56
00599*57
00600*57
IF (LC.NE.N16) GO TO 755 00601*57
00602*57
00603*57
GO TO (790,790,755,755,755),NDYN 00604*57
790 NDYN=4 00605*57
NCTCH=1 00606*56
00607*57
00608*57
00609*56
00610*56
CALL KSTAR (D(N4),D(NK15),D(N6),D(N7),NEQ,MBAND,NEQH,MBANDH,D(N8), 00611*56
00612*73
*D(N09),D(N010),D(N011),NWA,NEQB,NBLOCK,MI,D(N1),D(NPO12),D(ND012))00613*73
00614*53
755 CONTINUE 00615*26
00616*56
00617*98
00618*77
00619*77
CALL GETIME(ITIM9) 00620*80
TIM9=FLOAT(ITIM9)/1000. 00621*80
00622*77
00623*77
***** SOLVING EQU. OF MOTION FOR ALL CASES ***** 00624*77
00625*77
00626*77
SUM3=SUM3+(TIM9-TIM8) 00627*78
00628*78
00629*77
00630*77

```


8000 CONTINUE	00631*26
	00632
	00633
	00634
	00635
	00636
GO TO (780,760,750,780),KKK	00637*99
760 CONTINUE	00638*26
CALL GETIME(I1IM10)	00639*78
TIM10=FLOAT(I1IM10)/1000.	00640*80
	00641*80
	00642*76
	00643*76
***** STRAIN COMPATIBLE PROBLEM *****	00644*76
	00645*76
	00646*76
CALL STRAIN(D(NN07),NUMEL,NUMMAT,D(NN07),D(NN08),D(NN09),D(NN10),	00647*26
*D(NN11),D(NN12))	00648*28
	00649*46
CALL GETIME(I1IM11)	00650*29
TIM11=FLOAT(I1IM11)/1000.	00651*26
	00652*80
	00653*80
SUM4=SUM4+(TIM11-TIM10)	00654*78
	00655*78
	00656*78
	00657
	00658
GO TO 750	00659*30
***** CALL SHAKE AT THIS STAGE *****	00660
	00661*78
	00662*78
765 CONTINUE	00663*79
CALL GETIME(I1IM12)	00664*80
TIM12=FLOAT(I1IM12)/1000.	00665*80
	00666*78
	00667
780 CONTINUE	00668*30
	00669*81
	00670*77
CALL YLDCHK(FYLD,IPLNAX,SIG,LC,NUMEL,NPT)	00671*99
IF(KKS.EQ.1) GO TO 750	00672*99
	00673*99
	00674*99
	00675*77
	00676*77
500 FORMAT(415)	00677*90
WRITE(6,2009) IPT,IPT1,KK	00678*26
	00679*78
	00680*87
	00681*87
	00682*26
CALL SHAKE(NPK,NPJ,IPT,KK,D(N1),D(N4),D(N35),D(N36),D(N37),D(N39)	00683*78
*,N43,D(NK15),IPT1,D(N17),D(N029))	00684*78
	00685*89
	00686
GO TO 555	00687*43
768 CONTINUE	00688*79
	00689*78
	00690*78

```

CALL GETIME(TIM13)
TIM13=FLOAT(TIM13)/1000.
SUM5=SUM5+(TIM13-TIM12)

800 WRITE(6,2006)
900 GO TO 555
750 CONTINUE

600 CONTINUE
555 CONTINUE

WRITE(6,2010)
WRITE(6,2011) PINPUT,PELEMT,SUM,SUM1,SUM2,SUM3,SUM4,SUM5
505 FORMAT(15)
509 FORMAT(1H1/
* ' NUMBER OF LOAD CASES = /,15/
* ' NUMBER OF STRESS STATE IN THE BODY = /,15/
* ' NUMBER OF YIELD PLANES = /,15/
* ' NUMBER OF IIR. FOR KKS =2 = /,15/
* ' NCOND. FOR KKS =4 ; NCOND. =0;CENTROID = /,15/
* ' NTENS. FOR KKS=4 ; NTENS. =0;NO TENSION = /,15)

990 FORMAT(10A8)
1000 FORMAT(9I5)
1001 FORMAT(15,F10.0,15,F10.0,5I5)
1002 FORMAT(16I5)
1003 FORMAT(4F10.0,4I5)

**** ALL FORMAT RELATED TO PLOTTING *****
1008 FORMAT(5X,2I5)
1009 FORMAT(16I5)

1010 FORMAT(24I5)
1011 FORMAT(5X,2I5)
1012 FORMAT(5X,6I5)
1013 FORMAT(24I5)

***** FINISHED READING ALL FORMAT *****

2000 FORMAT(1H1//,8A8//
* ' NUMBER OF NODAL POINTS = /,15/
* ' NUMBER OF ELEMENTS = /,15/
* ' CODE FOR DYNAMIC OR STATIC ANALYSIS = /,15/
* ' EQ.1. LOAD VECTOR (DIRECT ANALYSIS) / /
* ' EQ.2. ARTHOUAKE ANALYSIS (DIRECT INTEGRATION) / /
* ' EQ.3. EIGENANALYSIS / /
* ' EQ.4. STATIC ANALYSIS (SOLID) / /
* ' EQ.5. TRANSIENT ANALYSIS ( SOLID FLUID ) / /
* ' CODE FOR PLANE PROBLEM ANALYSIY = /,15/
* ' EQ.-1. PLANE STRESS ANALYSIS / /

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```

00691*80
00692*80
00693*78
00694*78
00695*78
00696*78
00697
00698
00699
00700
00701
00702
00703*26
00704*26
00705*26
00706*31
00707*31
00708*95
00709
00710*77
00711*77
00712*77
00713*78
00714*01
00715*33
00716*33
00717*33
00718*33
00719*55
00720*76
00721*67
00722
00723*94
00724*46
00725*46
00726*46
00727*94
00728*94
00729*94
00730*46
00731*46
00732*46
00733*46
00734*46
00735*46
00736
00737*53
00738
00739
00740
00741
00742
00743
00744
00745
00746
00747**7

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* ' EQ.0. PLANE STRAIN ANALYSIS           ' /           00748**7
* ' EQ.1. AXISYMMETRIC ANALYSIS          ' /           00749**7
* ' NUMBER OF LOAD CASES                  = ' ,15/         00750
* ' NUMBER OF FREQUENCIES                  = ' ,15/         00751
* ' SOLUTION MODE (MODEX)                  = ' ,15/         00752
* ' EQ.0. FACETION                          ' /           00753
* ' EQ.1. DATA CHECK                       ' /           00754
* ' NUMBER OF SUBSPACE                      = ' ,15/         00755
* ' ITERATION CYCLES                        ' /           00756*75
2001 FORMAT(                               ' /           00757
* ' NUMBER OF CYCLES                        = ' ,15/         00758
* ' TIME STEP INCREMENTS                    = ' ,F10.3/       00759*14
* ' CODE FOR MATERIAL CONSTITUTIVE PROPERTIES = ' ,15/         00760*14
* ' EQ.1. ELASTIC ANALYSIS (ISOTROPIC)     ' /           00761*14
* ' EQ.2. STRAIN COMPATIBLE PROPERTIES (EQ. LINEAR) ' /         00762*14
* ' EQ.3. ELASTO PLASTIC ANALYSIS ( ZERO HARDENING ) ' /         00763*14
* ' EQ.4. SHARDEDEN ANALYSIS (ZERO HARDENING ) ' /         00764*20
* ' GRAVITATIONAL ACCL VALUE                 = ' ,E12.5/       00765*42
* ' TYPE OF ANALYSIS TO BE PERFORMED       = ' ,15/         00766*43
* ' EQ.0..... UNDRAINED ANALYSIS          ' /           00767*43
* ' EQ.1. UNDRAINED DRAINED ANALYSIS      = ' ,15/         00768*74
* ' BODY FORCES                             = ' ,15/         00769*74
* ' EQ.0 NOT INCLUDED                       ' /           00770*74
* ' EQ.1 INCLUDED                          ' /           00771*74
2002 FORMAT (I11/                          00772
* ' NUMBER OF BASE ACCEL POINTS             = ' ,15)         00773
2006 FORMAT (// ' REQUIRED DIMENSION OF ARRAY D EXCEEDS MTOT' ) 00774
2008 FORMAT (I11/                          00775
* ' TETA = ' , F10.3/                       00776
* ' BETA = ' , F10.3/                       00777
* ' GAMA = ' , F10.3/                       00778*77
* ' TOLERANCE FOR CONVERGANCE               = ' ,F10.6/       00779*77
* ' PUNCH THE OUTPUT E                      = ' ,15/         00780*77
* ' EQ.0. N3. PUNCH REQUIRED                  ' /           00781*77
* ' EQ.1. PUNCHING REQUIRED                   ' /           00782*77
* ' EQUILIBRIUM ITERATION NEEDED            = ' ,15/         00783*77
* ' EQ.0. NO                                ' /           00784*77
* ' EQ.1. YES                                ' /           00785*77
* ' NO.OF ITERATION IN TIME STEP MAX.      = ' ,15)         00786*77
* '                                          00787*77
* '                                          00788*77
* '                                          00789*77
2009 FORMAT(I11// ' TYPE OF OPTIMIZATION TO BE DONE = ' ,15/   00790*62
* ' EQ.1..... MINIMIZATION OF FUNCTION    ' /           00791*43
* ' EQ.2. .... MAXIMIZATION OF FUNCTION    ' /           00792*43
* ' EQ.3. .... MAXIMIZE FOR LIMIT ANALYSIS ' /           00793*81
* ' TYPE OF ANALYSIS                        = ' ,15/         00794*52
* ' EQ.0 MAXIMIZATION W. N. F. PLASTIC MULTIPLIERS ' /         00795*52
* ' EQ.1 MAXIMIZATION W.R.L. RESIDUAL STRESS ' /           00796*52
* ' BOUNDING TECHNIQUE TO BE APPLIED        = ' ,15/         00797*43
* ' EQ.0..... NONE                        ' /           00798*43
* ' EQ.1..... NEEDED                       ' /           00799*43
* '                                          00800
* '                                          00801*77
* '                                          00802*77
2010 FORMAT(I11//34H S O L U T I O N T I M E L O G //12X,
* ' 11HFOR PROBLEM//1X,12A6//)            00803*77
2011 FORMAT(49H INPUT PHASE .....F9.2// 00804*77
* ' 49H ELEMENT DATA AND STIFFNESS MASS LOAD .....F9.2// 00805*77
* ' 49H INPUT FOR STRAIN COMP. PROP. ....F9.2// 00806*77
* '                                          00807*78

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* 49H SOLID STIFFNESS MASS FREQUENCY CAL.F9.2// 00808*78
 * 49H ASSEMBLING OF STIFFNESS MASS LOAD ETC.F9.2// 00809*77
 * 49H SOLUTION MODES INCLUDING ALL TYPESF9.2// 00810*77
 * 49H STRAIN COMPATIBLE SOLUTIONF9.2// 00811*77
 * 49H SHANEDOWN SOLUTIONF9.2//)00812*77

00813
 00814
 00815

00816

STOP
 RETURN
 END

```

DATA SET MAXMIN AT LEVEL 012 AS OF 83/03/05
SUBROUTINE MAXMIN(STRESS,P1,P2,AG,BLKST,SHSTN,PMAX,PP,IPT,TIME,00001**3
*NUMEL,SM,SB,SHM)
IMPLICIT REAL*8(A-H,I-Z)
COMMON/MAINMOD/PTRESS(9),STRAIN(4),C(4,4),MPT,NEL
DIMENSION PMAX(NUMEL,1),STRESS(9)
STRESS(1)=STRESS(1)+PP
STRESS(2)=STRESS(2)+PP
STRESS(4)=STRESS(4)+PP
CC = (STRESS(1)+STRESS(2)) * 0.5
BB = (STRESS(1)-STRESS(2)) * 0.5
CK = DSQRT(BB**2 +STRESS(3)**2)
P1 = CC+CK
P2 = CC-CK
AG=45.0
IF(DABS(BB).LT.1.0E-6) GO TO 1
AG = 20.648*DATAHZ(STRESS(3),BB)
1 CONTINUE
STRESS(5)=P1
STRESS(6)=P2
STRESS(7)=(P1+P2)/2.
STRESS(8)=(P1-P2)/2.
COMPUTE BLKST,SHSTN ETC;
SM=(STRESS(1)+STRESS(2)+STRESS(4))/3.
DX=STRESS(1)-SM
DY=STRESS(2)-SM
DZ=STRESS(4)-SM
DS=STRESS(3)
SD=DSQRT(0.5*(DX*DX+DY*DY+DZ*DZ)+DS*DS)
IF(STRAIN(3).EQ.0) GO TO 5
SHM=(STRESS(3)/STRAIN(3))
5 CONTINUE
STRESS(9)=AG
2 CONTINUE
STRESS(4)=SM
STRESS(7)=SB
DO 3 J=1,8
BM=DABS(STRESS(J))
IF(BM-PMAX(NEL,J)) 180,100,100
100 PMAX(NEL,J)=BM
PMAX(NEL,9)=TIME

```

```

00002**9
00003**9
00004
00005
00006**7
00007**3
00008**9
00009**9
00010**9
00011
00012
00013
00014
00015
00016
00017
00018
00019**3
00020**3
00021**3
00022
00023
00024
00025**3
00026**4
00027**4
00028**3
00029**3
00030**3
00031**3
00032**3
00033**3
00034**3
00035**9
00036**9
00037**9
00038**9
00039**9
00040**9
00041**9
00042*11
00043*12
00044*12
00045*11
00046**3
00047**3
00048**3
00049**3
00050*10
00051**9
00052**9
00053**3
00054**3
00055**3
00056**3
00057**8
00058**8
00059**3

```

180 CONTINUE
3 CONTINUE

PF=PMAX(PF)
IF (PF=PMAX(NEL,9)) 95,200,200
200 PMAX(NEL,9)=PF
PMAX(NEL,18)=TIME
95 CONTINUE

RETURN
END

00060**3
00061**3
00062*12
00063*12
00064*12
00065*12
00066**3
00067**3
00068**3
00069**3
00070
00071
00072

DATA SET MDONEG AT LEVEL 001 AS OF 83/02/19
 SUBROUTINE MDONEG(ENEW,GNEW,PUIS,G)

IMPLICIT REAL *8(A-H,O-Z)

DIMENSION G(4,4)

100 DO 100 I=1,4
 DO 100 J=1,4
 G(I,J)=0.0

CONST=(1.-2.*PUIS)
 LAMBDA=(2.*PUIS*GNEW)/CONST
 C1=LAMBDA+2.*GNEW
 C2=LAMBDA
 C3=GNEW
 G(1,1)=C1
 G(1,2)=C2
 G(1,3)=G(1,2)
 G(3,3)=G(1,1)
 G(2,2)=C1
 G(2,3)=G(1,3)
 G(4,4)=C3

200 DO 200 I=1,4
 DO 200 J=1,4
 G(J,I)=G(I,J)
 CONTINUE

WRITE(17) ((G(I,J),J=1,4),I=1,4)

RETURN
 END

00001
 00002
 00003
 00004
 00005
 00006
 00007
 00008
 00009
 00010
 00011
 00012
 00013
 00014
 00015
 00016
 00017
 00018
 00019
 00020
 00021
 00022
 00023
 00024
 00025
 00026
 00027
 00028
 00029
 00030
 00031
 00032
 00033
 00034
 00035

```

DATA SET MIDEF      AT LEVEL 009 AS OF 83/02/11
SUBROUTINE MIDEF (TAU,DEPS,DP)
IMPLICIT REAL*8(A-H,O-Z)

      FORMS THE ELASTO-PLASTIC MATERIAL MATRIX

COMMON/EL/IND,ICORNI
COMMON /VMISES/A2,B2,C2,D2,A1,B1,C1,D1,YLD,BN,ISR,IST
      COMMON/MATHOD/STRESS(9), STRAIN(4),C(4,4),IP1,NEL
COMMON/CHORDA/WP
DIMENSION TAU(4),DEPS(4),DP(16)

      COMMON/SORIN/PI(5,12),XP(16),INDNB,ITYP2D

SM = (TAU(1)+TAU(2)+TAU(4))/3.
SX = TAU(1) - SM
SY = TAU(2) - SM
SS = TAU(3)
SZ = TAU(4) - SM
HPRIME=2.*D1/3.
BETA=1.5/YLD/YLD/(1.+HPRIME/A1)

IF (ITYP2D = 1) 10,11,12
10 WP= SA*DEPS(1) + SY*DEPS(2) + SS*DEPS(3) + SZ*DEPS(4)
   GU 10 15
11 WP= SA*DEPS(1) + SY*DEPS(2) + SS*DEPS(3)
   GU 10 15
12 DP1= A1 * (C1 - BETA*SX*SX)
   DP2= A1 * (C1 - BETA*SY*SY)
   DP3= A1 * (C1 - BETA*SS*SS)
   DP4= A1 * (D1 - BETA*SZ*SZ)

   DEPS(4)= (-DP1*DEPS(1)-DP2*DEPS(2)-DP3*DEPS(3))/DP4
   WP= SA*DEPS(1) + SY*DEPS(2) + SS*DEPS(3) + SZ*DEPS(4)

15 IF (WP.LT.0.) BETA=0.

DP ( 1) = A1 * (B1 - BETA*SA*SX)
DP ( 2) = A1 * (C1 - BETA*SA*SY)
DP ( 3) = A1 * (C1 - BETA*SA*SS)
DP ( 4) = A1 * (C1 - BETA*SA*SZ)
DP ( 5) = DP ( 2)
DP ( 6) = A1 * (B1 - BETA*SY*SY)
DP ( 7) = A1 * (C1 - BETA*SY*SS)
DP ( 8) = A1 * (C1 - BETA*SY*SZ)
DP ( 9) = DP ( 3)
DP (10) = DP ( 7)
DP (11) = A1 * (.5 - BETA*SS*SS)
DP (12) = A1 * (C1 - BETA*SZ*SS)

DP (13) = DP ( 4)

```

```

00001
00002
00003
00004
00005
00006**2
00007
00008**8
00009**4
00010
00011**4
00012**9
00013**9
00014
00015
00016
00017
00018
00019
00020
00021
00022
00023
00024
00025
00026
00027
00028
00029
00030
00031
00032
00033
00034
00035
00036
00037
00038
00039
00040
00041
00042
00043
00044
00045
00046
00047
00048
00049
00050
00051
00052
00053
00054**5
00055
00056

```



```

DP (14) = DP (8)
DP (15) = DP (12)
DP (16) = A1 * (B1 - BEJA*SZ*SZ)
00057
00058
00059
00060**5
00061

IF (ITYP2D.EQ.0.OR.ITYP2D.EQ.1) RETURN
00063
PRIME STRESS / MODIFY DE MATRIX
00064
DO 120 I=1,3
A=C(1,4)/C(4,4)
00065
00066
00067
00068
C(I,J)=C(I,J) - C(4,J)*A
00069
120 C(J,1) = C(1,J)
00070
IF (MP.LT.0.(6) DEFS(4)=DZ*(DEFS(1) + DEFS(2))
STRAIN(4)=STRAIN(4) + DEFS(4)
00071
00072
00073
00074
00075
00076
RETURN
END

```

DATA SET MUDCNV AT LEVEL 003 AS OF 82/03/16
 SUBROUTINE MUDCNV(NEL, IPT, D)

IMPLICIT REAL*8(A-H, O-Z)

DIMENSION D(4,4)

COMMON/NANCY/CM(12000)

COMMON/CONTI/DT, HED(8), KADN, NASIN, NOMEI, NEG, MBAND, NCYCL, NDYN, IPLNA

*X, ISTKPN, NPRINT, ISTART, NLEGN, MBANDH, NLEGS, MBANDS

00001
 00002
 00003
 00004
 00005
 00006
 00007
 00008
 00009

C

COMMON/HOBBY/KKS, NUMMAT, NC, LOOP, NDRN, NINT, IDWA, NPT

DO 5200 I=1,4

DO 5200 J=1,4

D(I,J)=0.0

5200 CONTINUE

NU=(NEL-1)*NPT*15+(IPT-1)*15

D(1,1)=CM(NU+1)

D(1,2)=CM(NU+2)

D(1,3)=CM(NU+3)

D(1,4)=CM(NU+4)

D(2,2)=CM(NU+5)

D(2,3)=CM(NU+6)

D(2,4)=CM(NU+7)

D(3,3)=CM(NU+8)

D(3,4)=CM(NU+9)

D(4,4)=CM(NU+10)

00010**2
 00011**2
 00012**2
 00013**2
 00014**2
 00015**2
 00016**2
 00017**2
 00018**2
 00019**2
 00020**2
 00021**2
 00022**2
 00023**2
 00024**2
 00025**2
 00026**2
 00027**2
 00028**2
 00029**2
 00030**2
 00031**2
 00032**2
 00033**2
 00034**2
 00035**2
 00036**2
 00037**2

DO 520 I=1,4

DO 520 J=1,4

520 D(J,I)=D(I,J)

IF(IPLNAX.NE.-1) GO TO 110

D(1,1)=D(1,1)-D(3,1)*D(1,3)/D(3,3)

D(1,2)=D(1,2)-D(3,2)*D(1,3)/D(3,3)

D(1,4)=D(1,4)-D(3,4)*D(1,3)/D(3,3)

D(2,2)=D(2,2)-D(3,2)*D(2,3)/D(3,3)

D(2,4)=D(2,4)-D(3,4)*D(2,3)/D(3,3)

D(4,4)=D(4,4)-D(3,4)*D(4,3)/D(3,3)

DO 130 I=1,4

DO 140 J=1,4

140 D(J,I)=D(I,J)

D(1,3)=0.0

D(3,1)=0.0

130 CONTINUE

110 CONTINUE

WRITE(6,102)

102 FORMAT(5X, ' PRINT IN MUDCNV.FOR')

WRITE(6,103) ((D(I,J),J=1,4),I=1,4)

103 FORMAT(5X,4E10.3)

DATA SET MODCUN AT LEVEL 020 AS OF 83/01/21
 SUBROUTINE MODCUN(C,NEL,PG,GSTRES)
 IMPLICIT REAL*8(A-H,O-Z)

COMMON/ NANCY/ CM(12000)
 COMMON/METHOD/STRESS(9),STRAIN(4),CC(4,4),IPT
 COMMON/PROPERTY/RS,NORMAT,NC,LOOP,NDRN,NINT,LDWA,NPT

DIMENSION C(4,4)

DIMENSION GSTRES(9)

NU=(NEL-1)*NPT*15+(IPT-1)*15
 CM(NU+1)=C(1,1)
 CM(NU+2)=C(1,2)
 CM(NU+3)=C(1,4)
 CM(NU+4)=C(1,3)
 CM(NU+5)=C(2,2)
 CM(NU+6)=C(2,4)
 CM(NU+7)=C(2,3)
 CM(NU+8)=C(4,4)
 CM(NU+9)=C(1,3)
 CM(NU+10)=C(3,3)

***** UPDATED STRESS VECTOR FOR RESIDUAL LOAD
 CALCULATION IN LOAD *****

CM(NU+11)=GSTRES(1)
 CM(NU+12)=GSTRES(2)
 CM(NU+13)=GSTRES(3)
 CM(NU+14)=GSTRES(4)
 CM(NU+15)=PG

101 WRITE(6,101)
 FORMAT(5A,' PRINT IN MODCUN.FOR ')
 WRITE(6,102)((C(1,J),J=1,4),I=1,4)
 102 FORMAT(5A,4E10.3)
 WRITE(6,104) (CM(1),I=1,15)
 104 FORMAT(2X,12E10.3)

RETURN
 END

00001*19
 00002
 00003**2
 00004**2
 00005**2
 00006*20
 00007*13
 00008**3
 00009**3
 00010**4
 00011**2
 00012**2
 00013*19
 00014**2
 00015**2
 00016*11
 00017*11
 00018*20
 00019*13
 00020*13
 00021*14
 00022*14
 00023**2
 00024**2
 00025**2
 00026**2
 00027**2
 00028**2
 00029**2
 00030**2
 00031**2
 00032**2
 00033**2
 00034**2
 00035**2
 00036**2
 00037**2
 00038*11
 00039*11
 00040*11
 00041*11
 00042*19
 00043**2
 00044*18
 00045*18
 00046**2
 00047
 00048

```

DATA SET NAMES          AT LEVEL 007 AS OF 83/02/24
SUBROUTINE NAMES (NEG,MBAND,NBLOCK,NEQB,NF,MTOT,IFPR,IFSS,RTOL,
*NITEM,COPU,NN15,NO15)
IMPLICIT REAL*(A-H,O-Z)
CALLS:  SECND, SBLCK, SSPCEB
CALLED BY:  SULEIG
PROGRAM TO COMPUTE SMALLEST EIGENVALUES AND ASSOCIATED VECTORS IN
THE GENERALIZED EIGENVALUE PROBLEM
      A*V=RT*B*V  (A POS DEF, B DIAG NONNEG DEF)
COMMON /SOL/ANORM, IDUM(5), NEIG, NAD, NVV, NFO
COMMON /TAPES/ NS1IF, NRFD, NL, NK, NT, NMASS, NUMBER
COMMON A(1)
COMMON /SWAM1/ N30, N31, N32, N33
      NS1IF=4
      NMASS=9
      NRFD=10
      NL=2
      NK=1
*****NT WAS ORIGINALLY 7 CHANGED BECAUSE OF WAT4 *****
*****CHECK THIS TAPE NUMBER AS AGAINST OF TAPE NO. 7*****
      N1=8
*****
PRINT EIGENPROBLEM SUMMARY
WRITE (6,1000) NEG,MBAND,NBLOCK,NEQB,NF
      IF (NEIG.GT.0) GO TO 300
D E T E R M I N A N T   S E A R C H
IF (NVV.GE.NF) GO TO 110
WRITE (6,1010) NF, NVV
STOP
110 CONTINUE
      N1M=3
      NVM=6
      NC=NF+N1M
      NCA=NEQ*MAX0(MBAND,NC)
*****CHANGED *****
      N1=NN15+1
      N2=N1+NCA
      N3=N2+NFO
      N4=N3+NEQ
      N5=N4+NFO
      N6=N5+NFO
      N7=N6+NEQ+NVM
      N8=N7+NEQ+NVM
      N9=N8+NC
      N10=N9+NC
      N11=N10+NC
      N12=N11+NC
      NO15=N12+NF

```

```

00001
00002**4
00003
00004
00005
00006
00007
00008
00009
00010
00011
00012
00013
00014**7
00015**3
00016
00017
00018
00019
00020
00021
00022
00023
00024
00025
00026**6
00027
00028
00029
00030
00031
00032
00033
00034
00035
00036
00037
00038
00039
00040
00041
00042
00043
00044
00045
00046**4
00047
00048
00049
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00052
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00054
00055
00056
00057
00058
00059**5

```

```

200 CALL SFCHID(A(N1),A(N2),A(N3),A(N4),A(N5),A(N6),A(N7),A(N8),A(N9)00060
1),A(N10),A(N11),A(N12),NEQ,MBAND,NF,NC,IFPR,ANOKM,CDFQ) 00061
GO TO 600 00062
SUBSPACE ITERATION 00063
300 NWA=NEQB*MBAND 00064
NV=2*NF 00065
IF (NF.GT.8) NV=NF+8 00066
IF (NAD.NE.0) NV=NAD 00067
IF (NVV.GE.NV) GO TO 310 00068
WRITE (6,1010) NV,NVV 00069
STOP 00070
310 NVV=NV*NEQB 00071
NTB=(MBAND-2)/NEQB+1 00072
IF (NTB.GE.NBLKCK) NTB=NBLKCK-1 00073
NVV=NVV*(NTB+1) 00074
CHECK FOR USE OF GIVEN STARTING ITERATION VECTORS 00075
IF (NFO.LE.0) GO TO 600 00076
REWIND 10 00077
READ (10) NEQ0,NBLK0,NEQB0,MBAND0,N10,NF0 00078
N2=1+NEQB0*NF0 00079
N3=N2+NEQB0*NV 00080
600 RETURN 00081
1000 FORMAT (/// 40H SOLUTION IS SOUGHT FOR FOLLOWING EIGENPROBLEM,// 00082
1 / 37H NUMBER OF EQUATIONS =,15 // 00083
2 37H HALF BANDWIDTH OF STIFFNESS MATRIX =,15 // 00084
3 37H NUMBER OF EQUATION BLOCKS =,15 // 00085
4 37H NUMBER OF EQUATIONS PER BLOCK =,15 // 00086
5 37H NUMBER OF EIGENVALUES REQUIRED =,15 // ) 00087
1010 FORMAT (/// 32H***ERROR SOLUTION TERMINATED, / 00088
1 12X,40HNUMBER OF NON-ZERO MASSES REQUIRED =, 15 / 00089
2 12X,40HNUMBER OF EXISTING MASSES IN THE MODEL =, 15 ) 00090
END 00091
00092
00093
00094
00095
00096
00097
00098
00099

```

DATA SET MULTI	AT LEVEL 001 AS OF 81/07/03	00001
SUBROUTINE MULTI (W,A,V,NN,MA)		00002
CALLED BY : SECVID		00003
IMPLICIT REAL*8(A-H,D-Z)		00004
DIMENSION A(1),W(1),V(1)		00005
		00006
NM=NN*(MA -1)		00007
NMA=NN - MA + 1		00008
DO 20 I=1,NN		00009
W(I)=0.0		00010
K=1		00011
IF (NMA -1) 10,15,15		00012
10 NM=NM - NN		00013
15 IL=NM + 1		00014
DO 20 J=1,IL,NN		00015
K=K + 1		00016
20 W(I)=W(I) + A(J)*V(K)		00017
		00018
IF (MA -1) 30,100,30		00019
		00020
30 KK=NN		00021
DO 40 I=2,MA		00022
II=I -1		00023
KK=KK + NN		00024
KJ=KK		00025
DO 40 J=1,II		00026
KJ=KJ - NN		00027
40 W(I)=W(I) + A(KJ + J)*V(J)		00028
IF (MA.EQ.NN) GO TO 100		00029
MAI=MA + 1		00030
IJ=1		00031
DO 50 I=MAI,NN		00032
KJ=KK		00033
IJ=IJ + 1		00034
II=I -1		00035
DO 50 J=IJ,II		00036
KJ=KJ - NN		00037
50 W(I)=W(I) + A(KJ + J)*V(J)		00038
		00039
100 RETURN		00040
END		00041
		00042

SUBROUTINE WRD01(M,A,B,XL,X)

DOUBLE PRECISION A,B,XL,X,SUMV
 DIMENSION A(1), b(1), XL(1), X(1)

K = 1
 DO 100 J=2,M
 L = M*(J-1)
 DO 100 I=1,J
 L=L+1
 K=K+1
 B(K)=b(L)

THE MATRIX B IS A REAL SYMMETRIC MATRIX.

MV=0
 CALL EIGEN (B,X,M,MV)

FORM RECIPROCAL OF SQUARE ROOT OF EIGENVALUES. THE RESULTS
 ARE PREMULTIPLIED BY THE ASSOCIATED EIGENVECTORS.

L=0
 DO 110 J=1,M

XL(J)=1.0/DSQR1(DABS(B(L)))

K=0
 DO 115 J=1,M
 DO 115 I=1,M
 K=K+1

B(K)=X(K)*XL(J)

FORM (B**(-1/2))PRIME * A * (B**(-1/2))

DO 120 I=1,M
 N2=0

DO 120 J=1,M
 N1=M*(I-1)
 L=M*(J-1)+1
 X(L)=0.0
 DO 120 K=1,M

N1=N1+1
 N2=N2+1
 X(L)=X(L)+b(N1)*A(N2)

L=0
 DO 130 J=1,M
 DO 130 I=1,J
 N1=I-M
 N2=M*(J-1)

L=L+1
 A(L)=0.0
 DO 130 K=1,M
 N1=N1+M
 N2=N2+1

A(L)=A(L)+X(N1)*B(N2)

COMPUTE EIGENVALUES AND EIGENVECTORS OF A

CALL EIGEN (A,A,M,MV)

L = 0
 DO 140 I=1,M

```
140 L=L+1
C XL(L)=A(L)
C
C COMPUTE THE NORMALIZED INTEGRATIONS
C
DO 150 I=1,M
N2=0
DO 150 J=1,M
N1=I-M
L=M*(J-1)+1
A(L)=0.0
DO 150 K=1,M
N1=N1+M
N2=N2+1
A(L)=A(L,J)+H(N1)*X(N2)
L=L+1
DO 180 J=1,M
SUMV=0.0
DO 170 I=1,M
L=L+1
SUMV=SUMV+A(L,J)*A(L)
SUMV=DSQRT(SUMV)
DO 180 I=1,M
K=K+1
X(K)=A(K)/SUMV
133 CONTINUE
RETURN
END
```



```

DATA SET 001          AT LEVEL 060 AS OF 83/03/05
SUBROUTINE 001(X1,X2,VEL,AC,AU,B,ID,DS,AMAX,PMAX,NUMNP,NUMEL, 00001
*NC,AN,FK,AN,ANI,LRD,IR,IK,NEU,KE,ITK) 00002*14
IMPLICIT REAL *8(A-H,I-Z) 00003*27
                                00004*14
                                00005
                                00006*14
                                00007**9
                                00008**9
                                00009
DIMENSION X1(1),X0(1),XZ(1),VEL(1),AC(1),AU(1),B(1),ID(NUMNP,1), 00010*47
*DS(4,3),AMAX(NUMNP,12,2),FMAX(NUMEL,1),AM(1),EK(1),AN(NNSC,1) 00011*11
*,ANI(NYI,1),LRD(1) 00012**9
                                00013*42
                                00014**9
                                00015*35
                                00016*32
                                00017*32
                                00018
COMMON/CONTR/DO,HED(8),RADN,NASIN,NUMSL,NKK,MBAND,NCYCL,NDYN,IPLNA 00019**3
*X,ISTPR,MPRINT,ISTART,NEGH,MBANDH,NEGB,MBANDS 00020*15
COMMON/PFTRK/AREA,WIDTH,HOR,VERT,ACCL 00021*46
COMMON/TAPES/NOU(6),NUMBER 00022*54
COMMON/LEVEL/WEIGHT,SHEAR 00023*54
                                00024*46
                                00025*45
                                00026*15

COMMON/BELA/DS,NELIYP,ND,NUNGT,MODEL,NCUN,IDW,NK015,NK016,NK017

***** PLATTING AND PRINTING ARRAY CHECK ? ***** 00027*15
COMMON/KAL1/JPT,REL,JPLUT(200),FELMT(200) 00028*15
COMMON/EL/IND,ICOUN1 00029*31
                                00030*30
                                00031*30
                                00032*27
                                00033*27
                                00034*28
COMMON/INTEG/ETA,BETA,GAMA,KTOL,NGAMA,NPUNCH,NEQUIB,ITEMX 00035
COMMON/ANIM/NTP,NSC,NSLC,NLC,NNYP,NNSC 00036*47
COMMON/PRIN/ISTPR,MPRINT,OPRINT,MPRINT,IP 00037**9
                                00038*46
                                00039**9

COMMON/GURA/IFLAG,IGR

COMMON/BLND1/DO(5,200,4,4),NCUN1

                                00040*19
                                00041*13
COMMON/HARVEI/DACC(700),DVEL(700),DDIDPD(700) 00042*56
                                00043*56
                                00044*22

WRITE(6,10)DS,NELIYP,ND,NUNGT,MODEL,NCUN,IDW,NK015,NK016,NK017
10  FORMAT(5X,10I5)

                                00045*22
                                00046*23
                                00047*23
                                00048*23
                                00049*13
                                00050*31
                                00051*18
***** BRANCH 001 FOR WORKING ARRAY INITIALIZATION IN ELT2D
IF ( IND.NE.-1) GO TO 200

```



```

*      5X,1      DISPL. PRINT(MPRINT) = ,15/      00105*47
*      5X,1      CONTOUR PRINT(JSIARI) = ,15/      00106*47
*      5X,1      COUNTER FOR STRESS = ,15/      00107*47
*      5X,1      COUNTER FOR DISPL. = ,15/      00108*47
*      5X,1      COUNTER FOR CONTOUR = ,15/      00109*47
*      5X,1      MODE OR ELEM PRINT(IP) = ,15/      00110*47
*      5X,1      CONVERGANCE FLAG(IFLAG) = ,15/      00111*47
*
TIME=DD*NC      00112**3
                00113**3
                00114
IF (NC)355,3333,355      00115*14
                        00116*26
                        00117*26
                        00118*14
                        00119*15
                        00120*15
355 CONTINUE      00121*14
                  00122*25
3333 CONTINUE      00123*14
IF (KKS.EQ.3.AND.IFLAG.EQ.0) GO TO 245      00124*29
IF (KKS.EQ.2.AND.LOOP.NE.NUMBER) GO TO 350      00125*29
                00126*29
                00127*59
                00128*59
                00129*59
IF (NPUNCH.NE.0) GO TO 351
WRITE(6,2017) TIME      00130*29
                00131*27
                00132*27
WRITE(6,2010) NC,ITR      00133*27
2010 FORMAT(/// 37H EQUILIBRIUM ITERATION IN TIME STEP = ,15//
*          37H NUMBER OF ITERATIONS = ,15 /)      00134*27
WRITE(6,2011)      00135*27
                00136
                00137*15
*****          DU LOOP STARTED HERE FOR ALL PRINTING AND      00138*15
*****          SAVING INFORMATION FOR ALL PLOTTING *****      00139*15
351 WRITE(65,991) TIME      00140*15
                00141*25
                00142*25
350 CONTINUE      00143*25
                00144*25
                00145*15
                00146
                00147*16
                00148*16
                00149*15
                00150*15
                00151
                00152
IP=1
DU 240 1=1,NUMNE      00153
                00154
                00155
DU 9900 1=1,4
DU 9900 1=1,3
9900 DS(1M,1J)=0.0
DU 235 J=1,4
K=ID(1,J)

```

	IF (N) 235, 235, 232	00156
232	DS(J,1)=b(K)	00157
	DS(J,2)=VEL(N)	00158
	DB(J,3)=AC(N)	00159
235	CONTINUE	00160
	IF (NC.EQ.0) GO TO 236	00161*36
	IF (MPRINT.LI.NPRINT) GO TO 241	00162*45
		00163*36
		00164*36
		00165*36
236	CONTINUE	00166*45
		00167*45
	IF (NPUNCH.EQ.0) WRITE(6,2012) 1,DS	00168*36
		00169
241	CONTINUE	00170
	DISP(1)=DS(1,1)	00171*37
	DISP(2)=DS(2,1)	00172*47
	DISP(3)=DS(3,1)	00173*48
	DISP(4)=DS(4,1)	00174*48
	DISP(5)=DS(1,2)	00175*48
	DISP(6)=DS(2,2)	00176*48
	DISP(7)=DS(3,2)	00177*48
	DISP(8)=DS(4,2)	00178*48
	DISP(9)=DS(1,3)	00179*48
	DISP(10)=DS(2,3)	00180*48
	DISP(11)=DS(3,3)	00181*48
	DISP(12)=DS(4,3)	00182*48
		00183*48
		00184*48
		00185*47
		00186*47
	***** DETERMINE MAXIMUM RESPONSE *****	00187*32
	DO 256 J=1,12	00188*32
	BB=DABS(DISP(J))	00189*32
257	IF (BB-AMAX(1,J,1)) 256,257,257	00190*32
	AMAX(1,J,1)=BB	00191*32
256	AMAX(1,J,2)=TIME	00192*32
	CONTINUE	00193*32
		00194*32
		00195*32
		00196*32
		00197*32
		00198*45
		00199*45
	IF (NC.EQ.0) GO TO 237	00200*45
	IF (MPRINT.LI.NPRINT) GO TO 240	00201*45
237	CONTINUE	00202*45
		00203*45
		00204*51
		00205*51
		00206*51
	IF (IP.GI.JPI) GO TO 240	00207*51
		00208*51
		00209*51
	LG=JPRINT(IP)	00210*32
	IF (LG.NE.1) GO TO 240	00211*32
		00212*32
		00213*50
	IF (KRS.EQ.2.AND.LOOP.NE.NUMBER) GO TO 240	00214*54

```

ANG=0.0
IF (DISP(1).EQ.0.0.AND.DISP(2).EQ.0.0) GO TO 15
00215*56
00216*56
00217*56
00218*50
00219*32
00220*46
00221*47
15 ***** CHECK ANG DOUBLE PRECISION *****
   ANG=KADN*DATAN2(DISP(1),DISP(2))
CONTINUE
00222*56
00223*56
00224*47
00225*46
2008 IF (NPUNCH.NE.0) WRITE(55,2008) J,HOK,VERT,DISP(1),DISP(2),ANG,
*DISP(3),DISP(4)
   FORMAT(13,7E11.4/)
00226*46
00227*46
00228*32
00229*52
00230*59
00231*60
00232*52
90   FORMAT(10I5)
00233*52
00234*52
00235*30
00236*50
238 CONTINUE
00237*50
00238*50
00239
00240
00241
IP=IP+1
240 CONTINUE
00242*15
00243
00244*15
***** CALL SUBROUTINE SAVLPD AT THIS STAGE
00246*15
00247*15
00249*14
00250*14
242 CONTINUE
***** STRESS OUTPUT PRINTING AND INFORMATION SAVING *
00251*36
00252*14
00253*14
00254*14
00255*14
3334 CONTINUE
00256
245 CONTINUE
00257*16
NNN=NC
00258
00259*15
***** STRESS OUTPUT ROUTINE TO COMPUTE STRESSES
***** ELEMENT NODAL DISPLACEMENTS *****
00260*15
00261*15
WRITE(6,10) ND,NELTYP,ND,NURGT,MODEL,NCUN,LDW,NK015,NK016,NK017
00262*15
CALL STRSPK(B,TIME,FMAX,NUMEL,NC,AM,EK,AN,AH1,ENJ,LC,ITR,RE)
00263*30
00264*30
00265*30
WRITE(6,10) NS,NELTYP,ND,NURGT,MODEL,NCUN,LDW,NK015,NK016,NK017
00266*27
00267*15
00268*15
00269*17
00270*25

```

***** CHECK FOR ISFEES CONNIFE *****

00271*25
00272*25
00273*29
00274*29
00275
00276
00277*32
00278
00279
00280**8
00281*29
00282*29
00283*58
00284*29
00285*29
00286
00287**8

250 CONTINUE

600 CONTINUE

IF(NC.EQ.0) GO TO 800

IF(KKS.EQ.3.AND.IFLAG.EQ.0) GO TO 700

800 CONTINUE

IF(MPRINT.LI.NPRINT) GO TO 850

MPRINT=0

GO TO (850,850,850,850),KKS

00288*31
00289*31
00290*30

~~850 CONTINUE~~

IF(KKS.EQ.3.AND.NEQUIF.NE.0) GO TO 855

IF(KKS.EQ.4.AND.NCUND.LE.1) GO TO 855

CALL ASSEFF(FE,NUMEL,NEQ)

00291*33
00292*33
00293*33
00294*34
00295*39
00296*39
00297*39

WRITE(0,857)
857 FORMAT(1H1/5X,' PRINT OF RESIDUAL LOAD VECTOR IN OUT AT EACH
+ LOAD STEP *****')

WRITE(0,858) (RF(I),I=1,NFO)
858 FORMAT(5X,8F12.5)

00298*34
00299*34
00300*39
00301*33
00302*30
00303*31

REWIND 21

~~855 CONTINUE~~

00304*33
00305*31
00306*30
00307*36
00308*36
00309*19
00310
00311

NC=NC+1

IF(NC-NCYCL) 700,900,900

900 CONTINUE

IF(KKS.EQ.4) GO TO 700

```
WRITE(0,2020) (1,(EMAX(I,U),U=1,18),I=1,NUMBER)
00312**3
00313*45
```

```
IF(NPUNCH.NE.0) WRITE(47,2030) (1,(EMAX(I,U),U=1,9),I=1,NUMBER)
```

```
2030 FORMAT(15,8E9.2,3X/E9.2/)
```

```
WRITE(0,2023)
00314*49
00315*49
00316*45
DO 750 I=1,NUMBER
00317*47
DO 750 K=1,2
00318*47
WRITE(0,2021) J,(AMAX(I,U,K),U=1,12)
00319*47
```

```
2031 IF(NPUNCH.NE.0) WRITE(40,2031) 1,(AMAX(I,U,1),U=1,12)
FORMAT(13,7E11.4/5E10.3/)
```

```
750 CONTINUE
00320*47
```

```
IF(NRS.EQ.2.AND.LOOP.NE.NUMBER) GO TO 700
00321*47
00322*50
```

```
***** PUNCH ON CARD FOR PLOTTING *****
00323*57
00324*57
```

```
IF(NPUNCH.EQ.0) GO TO 700
00325*32
00326*41
```

```
NTERM=(NCYCL/NPRINT)+1
00327*45
NSTRT=(NCYCL/101TRK)+1
00328*45
NCUNT=(NCYCL/151AKT)+1
00329*45
00330*45
```

```
WRITE(0,13) NTERM,NSTRT,NCUNT
00331*45
00332*45
00333*45
00334*46
00335*46
00336*46
00337*46
```

```
13 FORMAT(1H1/5X,' DO OF DISPL. PRINT REQUEST = ',15/
* 5X,' DO OF STRESS PRINT REQUEST = ',15/
* 5X,' NO. OF STRESS CONTR.REQUEST = ',15/)
```

```
IF(NPUNCH.NE.0) GO TO 700
00338*46
00339*46
00340*46
00341*45
00342*41
```

```
REWIND 55
00343*41
```

```
REWIND 57
00344*41
```

```
REWIND 18
00345*41
```

```
REWIND 47
```

```
REWIND 46
```

```
80 WRITE(0,80)
FORMAT(1H1/' ***** PRINT DISPL HISTORY FOR PUNCH **'/)
00346*52
00347*52
00348*52
00349*52
```

```
DO 710 I=1,NTERM
00350*41
00351*45
00352*45
```

```
STEP=(I-1)*NPRINT+1
```

```
READ(05,991) TIME
```

```
WRITE(0,991) TIME
```

```
991 FORMAT(1X,' TIME STEP FOR DISPL. OUTPUT = ',F10.4)
```

	DO 710 K=1, JPI	00353*57
	READ(55, 2008) I, HOK, VENT, DISP(1), DISP(2), ANG, DISP(5), DISP(9)	00354*46
	WRITE(6, 9) I, HOK, VENT, DISP(1), DISP(2), ANG, DISP(5), DISP(9)	00355*46
710	CONTINUE	00356*46
		00358*41
		00359*41
2022	FORMAT(1H1/5X, 13, 12E10.3)	00360*41
		00361*42
7	FORMAT(5X, 13, 8E10.3)	00362*45
8	FORMAT(15, 0E9.2, 5X)	00363*52
12	FORMAT(13, 7E11.4/8E10.3/)	00364*54
		00365*54
9	FORMAT(13, 7E11.4/)	00366*54
		00367*46
		00368*45
		00369*44
		00370*52
		00371*52
85	WRITE(6, 85)	00372*52
	FORMAT(1H1/' ***** PRINT OF STRESS HISTORY ***'/)	00373*52
		00374*45
	DO 715 N=1, NSTRP	00375*45
	STEP=(N-1)*10+1+10	
	READ(57, 992) TIME	
	WRITE(6, 992) TIME	
992	FORMAT(1X, ' TIME STEP FOR STRESS OUTPUT = ', F10.4/)	
		00376*44
	DO 715 K=1, NEL	00377*43
		00378*43
		00379*45
		00380*46
	READ(57, 2032) NL, (SIG(1), I=1, 2), SIG(4), (STRAIN(1), I=1, 3),	00381*46
	* SIG(5), SIG(6), SIG(8), SH, SD, PG, SHM, EPSV, WF	00382*46
		00383*46
2032	FORMAT(13, 7E11.4/8E10.3/)	00384*46
	WRITE(6, 11) NL, (SIG(1), I=1, 2), SIG(4), (STRAIN(1), I=1, 3),	00384*46
	* SIG(5), SIG(6), SIG(8), SH, SD, PG, SHM, EPSV, WF	00385*46
		00386*46
		00389*44
715	CONTINUE	00390*42
		00391*46
11	FORMAT(13, 7E11.4/8E10.3/)	00392*46
		00393*55
		00394*53
95	WRITE(6, 95)	00395*53
	FORMAT(1H1/' ***** PRINT OF STRESS CONTOUR ***'/)	00396*46
		00397*45
	DO 730 N=1, NCONT	
	STEP=(N-1)*4+1+4	
	READ(18, 992) TIME	
	WRITE(6, 992) TIME	
993	FORMAT(1X, ' TIME STEP FOR CONTOUR OUTPUT = ', F10.4/)	
	DO 730 K=1, NUMFL	00398*52
		00399*45
	READ(18, 2033) NL, SIG(1), SIG(2), SIG(4), SIG(8), PG, SHM, EPSV, STRAIN(00400*46	
*4)		
2033	FORMAT(15, 8E9.2, 5X)	00401*46
	WRITE(6, 7) NL, SIG(1), SIG(2), SIG(4), SIG(8), PG, SHM, EPSV, STRAIN(4)	00402*46
		00403*46

		00405*47
		00406*46
		00407*46
		00408*45
730	CONTINUE	00409*45
		00410*42
98	WRITE(6,98) FORMAT(1H1/' *** PRINT OF MAX. DISPL. '/) DO 740 K=1,NUMEL KD=K READ(6,2031) (KD,(AMAX(K,J),J=1,12)) WRITE(6,2031) (KD,(AMAX(K,J),J=1,12)) CONTINUE	
740		
99	WRITE(6,99) FORMAT(1H1/' *** PRINT OF MAX. DISPL. VEL ACCN. '/) DO 745 K=1,NUMEL KD=K READ(6,2030) (KD,(VMAX(K,J),J=1,9)) WRITE(6,2030) (KD,(VMAX(K,J),J=1,9)) CONTINUE	
745		
700	CONTINUE	00411*25
		00412*44
		00413*44
		00414*44
		00415*25
	RETURN	00416
		00417
		00418
1009	FORMAT(315)	00419
1010	FORMAT(15,F10.0)	00420
		00421
2011	FORMAT (00422
	* 4X, 'NUMBER', 17A, 'DISPLACEMENT', 30X, 'VELOCITY', 30X, 'ACCELERATION' /	00423
	* 4X, 'NUMBER', 5X, 'UR', 5X, 'UZ', 5X, 'WR', 5X, 'WZ', 8X, 'UR', 5X, 'UZ', 5X,	00424
	* 'WR', 5X, 'WZ', 8X, 'UR', 5X, 'UZ', 5X, 'UZ', 5X, 'WR' //)	00425
2012	FORMAT(15,12E10.3)	00426
2016	FORMAT(1H1/' NODAL POINT LOADS AT TIME = ',F10.3//)	00427
2017	FORMAT(1H1/' RESPONSE AT TIME = ',F10.5//)	00428
2018	FORMAT(1H1/' INITIAL CONDITIONS '//)	00429
2020	FORMAT(1H1/10X,' MAX. STRESSES ' /	00430
	* ELEMENT',NUMBER',5X,'SIGX',5X,'SIGZ',5X,'SIGT',5X,'SIGRZ',5X,'S	00431
	*IGMAX',4X,'SIGMIN',5X,'MEAN NORMAL',5X,'MAX. DEVIATOR',4X,'F'/1000	00432**3
	* (16,9E10.3 /6X,9E10.3//)	00433**3
2023	FORMAT(1H1/' PRINT OF DISPL.,VEL.,ACCN., MAXIMA TIME '//)	00434*49
2021	FORMAT(5X,16,12E10.3//)	00435*49
		00436*49
		00437*32
	END	00438

```

DATA SET OUTPUT      AT LEVEL 020 AS OF 83/02/24
SUBROUTINE DD1F01(NUMEL,NIP,NBC,NPJ,NPK,FSOL,SG,AN,SD,SN,CT,SU,
+NEWS,IN,NDIRNP)    00001*10
                                00002*10
IMPLICIT REAL * 8(A-H,O-Z)    00003
                                00004
                                00005
DIMENSION PSOL(1),AN(NPJ,1),SD(NPJ,1),EP(3),ANI(14,3),AP(14
*) ,SN(NPJ),IN(NDIRNP,2)    00006**9
                                00007*17
                                00008**9
                                00009**8
DIMENSION ST(20,12),T1(5,12),F(4,8),LM(10),PS(10,6),FACN(4)    00010*12
DIMENSION SP(3),G(2)    00011**8
DIMENSION CT(NPJ,1),SU(1),RM(5),ZM(5)    00012*10
DIMENSION LP(4)    00013*13
                                00014**8
                                00015**8
COMMON/GRAND/COH(20),PHI(20),ITYPE(20)    00016*20
COMMON/INTEGE/IEIA,BETA,GAMA,RTOL,NGAMA,NPUNCH,NEQUIB,ITEMX    00017*20
                                00018
COMMON/BLNDT/SM(5,200,*,*),NCOND    00019
                                00020**5
COMMON/JAKA/LPC,N,M1,M2,NPT,KPT,PERKT,MD,MM    00021*11
                                00022*11
COMMON/BOBBY/RRS,NUMGT,RC,LOOP,NDKN,NINT,LDWA,NGT    00023*10
                                00024*10
REWIND 10
REWIND 55
NDT=1
IF(NCOND.EQ.1) NDT=4
                                00025**8
                                00026**8
WRITE(6,500)    00027**8
500 FORMAT(1H) ' **** PRINT OF SOLUTION VECTOR FOR PROBLEM ' )    00028**8
                                00029**8
WRITE(6,600) (FSOL(I),I=1,N)    00030**9
600 FORMAT(5X,10E12.5)    00031**8
                                00032**8
                                00033**8
PI=4.0*DATAFAN(1.0D0)    00034**8
RADN=180./PI    00035**8
RAD2=0.5*RADN    00036**8
REWIND 11    00037**8
                                00038**8
                                00039**8
WRITE(6,400)    00041
                                00042**5
IF(KPT.EQ.2.AND.NPT.EQ.1.OR.KPT.EQ.2.AND.NPT.EQ.2) GO TO 350
                                00045
WRITE(6,700)
DO 32 I=1,NUMEL    00046**3
WRITE(6,609)I    00047
                                00048*10
                                00049*10
                                00050*10
DO 35 J=1,NIP    00051*11
                                00052*12

```

00044

DO 33 I=1,NPT	00053*12
LX=J+(I-1)*NYP*NDI+(I-1)*NYP	00054*10
	00055*10
	00056*10
33 LP(LP)=LX	00057*12
CONTINUE	00058*12
	00059*12
IF(NCOND.NE.1) GO TO 36	
WRITE(6,608)J,PSOL((LP(1)+1)),PSOL((LP(2)+1)),	00060*12
*PSOL((LP(3)+1)),PSOL((LP(4)+1))	00061*12
GO TO 35	
36 WRITE(6,608) 0,PSOL((LP(1)+1))	00062*11
35 CONTINUE	00063*10
32 CONTINUE	00064*10
	00065
	00066
WRITE(6,698)	00067
DO 36 M=1,NUMEL	00068**3
DO 37 I=1,NDI	00069*10
	00070
***** RE TRANSFER AN ARRAY TO ANI ARRAY *****	00071
DO 200 J=1,NYP	00072
LX=J+(M-1)*NYP*NDI+(I-1)*NYP	00073*10
DO 100 I=1,NSC	00074
LY=I+(M-1)*NSC*NDI+(I-1)*NSC	00075*10
100 ANI(J,I)=AN(LY,LX)	00076
200 CONTINUE	00077
	00078
DO 34 K=1,NSC	00079
EP(K)=0.0	00080
DO 30 J=1,NYP	00081
AP(J)=ANI(J,K)	00082
LX=J+(M-1)*NYP*NDI+(I-1)*NYP	00083*10
30 EP(K)=EP(K)+AP(J)*PSOL(LX+1)	00084
IF(EP(K).NE.0.0) GO TO 34	
LY=K+(M-1)*NSC*NDI+(I-1)*NSC	
DO 31 N=1,APJ	
ANIT(N,LY)=0.0	
ANIT(LY,N)=0.0	
31 CONTINUE	
34 CONTINUE	
WRITE(6,699)M,(EP(K)),I=1,NSC	00085
IF(NPUNCH.EQ.0) GO TO 39	00086*20
IF(LP.NE.1) GO TO 39	00087*20
WRITE(18,40) M,(EP(K)),K=1,NSC	00088*20
39 CONTINUE	00089*20
	00090*10
37 CONTINUE	00091*10
36 CONTINUE	

***** PRINT HERE DISPLACEMENT *****

IF (KPT.EQ.2.AND.NPT.EQ.0) GO TO 300

FORMAT(215)

WRITE(6,301)

DO 303 I=1,NUMNP

DO 302 J=1,2

U(J)=0.0

LL=LN(1,U)

IF(LL) 308,308,305

U(J)=PSOL(LL+1+NRK)

CONTINUE

CONTINUE

WRITE(6,304)1, (U(K),K=1,2)

IF(NP.ONCH.NE.0) WRITE(65,304) 1,U(1),U(2)

CONTINUE

IF(KPT.EQ.1) GO TO 1000

00092*10

K1=(NSC*1-2)+(M-1)+NSC*NGT

K2=NSC*1+(M-1)*NSC+NGT

SN(K1)=FP(1)

SN(K2)=FP(2)

SN(K3)=FP(3)

00093*10

00095

00096

00097**5

300 CONTINUE

301 FORMAT(1H17/001P1 TABLE. . . NODAL DISPLACEMENTS //

*13X,4HNODE,9X,11H0 = X-DISP.,9X,11HV = Y-DISP./)

304 FORMAT(5X,15,ZE10.3)

WRITE(6,695)

DO 265 I=1,NPJ

SUM=0.0

DO 285 J=1,NPK

285 SUM=SUM+SD(1,U)+PSOL(J+1)

SN(1)=SUM

265 CONTINUE

00098**3

00100

00101

00102

00103

00104

00105

00106**5

GO TO 370

00107**5

00108**5

00109**5

00110**5

00111**5

00112**5

00113**5

350 DO 360 I=1,NPJ

360 SN(1)=PSOL(I+1)-PSOL(I+1+NPJ)

370 CONTINUE

WRITE(6,612)

DO 290 M=1,NBREL

00114**5

00115**3

00116*10

DO 295 I=1,NPJ

00117*10

00118*11

```

K1=(NSC+1-2)+(M-1)*NSC*N1/1
K2=NSC+1+(M-1)*NSC*N1/1
WRITE(6,699)M,(SN(K),K=K1,K2)
IF(MFON(1,FO,0) GO TO 49
IF(JF,PF,1) GO TO 49
WRITE(7,40) M,(SN(K),K=K1,K2)
49 CONTINUE
40 FURMAI(15,3E10,3)

295 CONTINUE
290 CONTINUE
CALL CHECK(CY,SH,SO,NFO,MEGS)
GO TO 1000

***** CONVERT STRESSES TO RADIAL COORDINATES *****
WRITE(6,695)
DO 800 M=1,NUMEL
READ(11) S1,P1,P,DM,ALFA,EM,MM,ZM,RS,RT,IMAT,DRR,EMU,YOUNG,POIS,
* VOL,PS,FACH
DO 900 I=1,NM1
K1=(NSC+I-2)+(M-1)*NSC*ND1
K2=NSC+1+(M-1)*NSC*ND1
SP(1)=SH(K1)
SP(2)=SN(K1+1)
SP(3)=SN(K1+2)
THETA=DATAN(ZM(1)/RM(1))
HARV=DCOS(THETA)
DES=DSIN(THETA)
SARTU=DSIN(2.*THETA)
SN(K1)=SP(1)*(HARV**2.)+S1(2)*(DES**2.)+SP(3)*(SARTU)
SN(K1+1)=SP(1)*(DES**2.)+SP(2)*(HARV**2.)-SP(3)*(SARTU)
SN(K1+2)=(SP(2)-SP(1))/2.*(SARTU)+SP(3)*(DCOS(2.*THETA))
RZ=DSQR1(RM(1)**2.+ZM(1)**2.)
PAL=(Z.*(COSH(IMAT)))
PAL2=2.*PAL
SN(K1)=SN(K1)/PAL
SN(K1+1)=SN(K1+1)/PAL
SN(K1+2)=SN(K1+2)/PAL

```

```

00119*10
00120*10
00121*10
00122
00123*20
00124*20
00125*20
00126*20
00127*20
00128*10
00129*10
00130*10
00131*10
00132
00133**8
00134**12
00135**8
00136**8
00137**8
00138**8
00139**8
00140**8
00141*10
00142**8
00143**8
00144**8
00145**8
00146*10
00147*10
00148**8
00149**8
00150*10
00151*10
00152**8
00153**8
00154**8
00155**8
00156**8
00157**8
00158**8
00159*12
00160**8
00161**8
00162**8
00163**8
00164**8
00165**8
00166**8
00167*17
00168*17
00169*17
00170*17
00171*17
00172*17
00173*17
00174*17
00175*17

```



```

DATA SET PRAGER AT LEVEL 008 AS OF 83/02/11
SUBROUTINE PRAGER (TAU,DEPS,DP)
IMPLICIT REAL*8(A-H,O-Z)
00001
00002
00003
FORMS THE ELASTO-PLASTIC MATERIAL MATRIX
(FOR THE DRUCKER-PRAGER FIELD CRITERION)
00004
00005
00006
COMMON/EL7/IND,ICORNT
00007**2
COMMON/MATMOD/STRESS(9), STRAIN(4),C(4,4),JPT,NEL
00008**7
COMMON /DKPRAG/ A1,B1,C1,A2,B2,C2,D2,G,BN,ALFA,YLD,ISR,ISI
00009
00010
COMMON/SORIN/IT(5,12),AP(16),INDNH,ITYP2D
00011**8
DIMENSION TAU(4),DEPS(4),DP(16)
00012**8
00013**2
00014
00015
SM = (TAU(1)+TAU(2)+TAU(4))/3.
00016
SX = TAU(1) - SM
00017
SY = TAU(2) - SM
00018
SZ = TAU(4) - SM
00019
SBAR=DSQRT( .5*(SX*SX+SY*SY+SZ*SZ) +SS*SS )
00020
SBAR=DSQRT (G +9.*BN*ALFA*ALFA)
00021
00022
AA=G/SBAR/QTQ
00023
BB=3.*BN*ALFA/QTQ
00024
00025
IF (ITYP2D - 1) 10,11,12
00026
10 DLAMDA= BB * (DEPS(1)+DEPS(2)+DEPS(4)) +
00027
1 AA * (SX*DEPS(1)+SY*DEPS(2)+SS*DEPS(3)+SZ*DEPS(4))
00028
GO TO 15
00029
00030
11 DLAMDA= BB* (DEPS(1)+DEPS(2)) +
00031
1 AA * (SX*DEPS(1)+SY*DEPS(2)+SS*DEPS(3))
00032
GO TO 15
00033
00034
12 SA= AA*SX + BB
00035
SB= AA*SY + BB
00036
SC= AA*SS
00037
SD= AA*SZ + BB
00038
00039
DP1= B2 - SA*SD
00040
DP2= B2 - SB*SD
00041
DP3= - SC*SD
00042
DP4= A2 - SD*SD
00043
00044
DEPS(4)=(-DP1*DEPS(1)-DP2*DEPS(2)-DP3*DEPS(3))/DP4
00045
00046
DLAMDA= BB * (DEPS(1)+DEPS(2)+DEPS(4)) +
00047
1 AA * (SX*DEPS(1)+SY*DEPS(2)+SS*DEPS(3)+SZ*DEPS(4))
00048
00049
IF (DLAMDA.LT.0.) GO TO 10
00050
SX=SA
00051
SY=SB
00052
SS=SC
00053
SZ=SD
00054
GO TO 25
00055
00056
15 IF (DLAMDA.GT.0.) GO TO 20
00057
16 AA=0.
00058
BB=0.
00059

```

```

DATA SET PRINTED AT LEVEL 003 AS OF 83/02/16
SUBROUTINE PRINTD(IH,F,B,REQB,NUMBP,LD,NBLOCK,RFQ,NT,MQ)
IMPLICIT REAL*8(A-H,O-Z)
00001**2
00002
00003
00004
00005
00006**3
00007**2
00008
00009
00010
00011
00012
00013
00014
00015
00016
00017
00018
00019
50 01=Q12
00020
02=Q22
00021
00022
00023
01=Q13
00024
02=Q23
00025
REWIND N1
00026
READ (N1)
00027
00  WRITE (6,2003) Q1,Q2
00028
00029
00030
00031
00032
00033**2
00034**2
00035
100 F(I,I)=0.
00036
IF (M.GT.NN) GO TO 150
00037
IF (M.EQ.0) GO TO 150
00038
READ (N1) B
00039
NN=NN-REQB
00040
150 IF (IH(I,1).LT.1) GO TO 250
00041**2
K=M-NN
00042
M=M-1
00043
00044
00045
DO 200 I=1,LI
00046
200 F(I,I)=B(K,I)
00047
250 I=I-1
00048
00049**2
WRITE(6,2004) N,(L,(F(I,I),I=1,2),I=1,LI)
00050
500 N=N-1
00051
00052
RETURN
00053
00054
2003 FORMAT (1H1,30H1) D I S P L A C E M E N T S /
00055
1 17H U T A 1 1 U N S, / 3X,4HNODE,2X,A6,2(12X,2HX-,12X,
00056
2 2HY=,12X,2HZ=), / 7H NUMBER,2X,A6,3(3X,11HTRANSLATION),
00057
3 3(6X,6HROTATION), / 1X)
00058
2004 FORMAT(1H1,10,18,2E14.5 / (7X,18,2E14.5) )
00059**3

```



```

DATA SET PRP1 AT LEVEL 010 AS OF 82/03/16
SUBROUTINE PRP1(N,PRP,NUMMAT)
IMPLICIT REAL*8(A-H,O-Z)
COMMON/DELA/MS,RELTFF,ND,NUMST,MODEL,NCUN,LDW,NK015,NK016,NK017
00001**7
00002
COMMON/BBBY/KKS,NUMGAT,IC,LDUP,NDRN,NINT,LDWA,NPT
00003**4
00004*10
00005*10
00006*10
00007
00008
DIMENSION PRP(1)
00009**4
00010
00011
00012**4
00013**4
00014**4
IF(N.P0.1) WRITE(6,2000) MODEL,NUMMAT,NCUN,LDW
*,NINT
00015*10
00016*10
00017**4
00018**4
00019**4
00020**4
GO TO (1,2),MODEL
00021**4
***** MODEL = 1 ELASTOPLASTIC (VON M
1 WRITE(6,2106) (PRP(I),I=1,NCUN)
00022**4
00023**4
00024**4
00025**7
GO TO 3
***** MODEL = 2 ELASTOPLASTIC (DRUCKER PRAGER)
00026**4
00027**4
2 WRITE(6,2107) (PRP(I),I=1,NCUN)
00028**4
00029**4
3 CONTINUE
00030**7
00031**4
00032**4
00033**6
2000 FORMAT(1H1/' MATERIAL DEFINATION = ',15/
* ' EQ.1. ELASTOPLASTIC ANALYSIS (VON MISSES)'
00034**6
* ' EQ.2. ELASTOPLASTIC ANALYSIS (DRUCKER PRAGER)
00035**6
* ' )'
00036**6
* '
00037**6
* ' NUMBER OF DIFFERENT MATERIAL SETS = ',15/
00038**6
* ' NUMBER OF CONSTANTS SETS PER MATERIAL
00039**6
* SETS = ',15/
00040**6
* ' DIMENSION OF STORAGE ARRAY FOR CENTER
00041**6
* STRESSES = ',15/
00042*10
* ' NUMBER OF INTEGRATION POINT = '15//)
00043*10
00044**4
2106 FORMAT(1H,4X,29HE .....( PRP(1) ).. =, E14.6/,
* 1H,4X,29HVND .....( PRP(2) ).. =, E14.6/,
* 1H,4X,29HVYIELD .....( PRP(3) ).. =, E14.6/,
* 1H,4X,29HV (HARDEN) .....( PRP(4) ).. =, E14.6///)
00048**4
2107 FORMAT(1H,4X,29HE .....( PRP(1) ).. =,E14.6/,
* 1H,4X,29HVND .....( PRP(2) ).. =, E14.6/,
* 1H,4X,29HVCONSTITION .....( PRP(3) ).. =, E14.6/,
* 1H,4X,29HV FRICTION ANGLE.(PRP(4) ).. =, E14.6///)
00051**4
00052**4
00053**4
00054**4
RETURN
00055
END
00056

```

```

DATA SRT PRSTS      AT LEVEL 014 AS OF 83/02/25
SUBROUTINE PRSTS(STRESS,PG,GSTRES)
IMPLICIT REAL*8(A-H,O-Z)

```

```

00001**9
00002
00003**5
00004*10
00005*12
00006*12
00007*10
00008**5
00009*11
00010*11
00011**5
00012**5
00013*10
00014**2
00015**2
00016*14
00017**2
00018**2
00019**2
00020**9
00021**9
00022**9
00023**4
00024**2
00025*13
00026*10
00027*10
00028*10
00029*10
00030*10
00031*10
00032*10
00033*10
00034*10
00035*10
00036**2
00037**9
00038**3
00039

```

```

COMMON/HORBY/KKS,NUMPAT,NC,LOOP,WDRN,NINT,LDWA,NPT

```

```

COMMON/MATMOD/PTRES(9),STRAIN(4),C(4,4),IPT

```

```

DIMENSION GSTRES(9),STRESS(9)

```

```

IF(WDRN.NE.0) GO TO 100

```

```

GSTRES(1)=STRESS(1)+PG
GSTRES(2)=STRESS(2)+PG
GSTRES(3)=STRESS(4)+PG
GSTRES(4)=STRESS(3)

```

```

GO TO 200

```

```

100 CONTINUE

```

```

DO 150 J=1,2

```

```

150 GSTRES(1)=STRESS(1)

```

```

GSTRES(3)=STRESS(4)

```

```

GSTRES(4)=STRESS(3)

```

```

200 CONTINUE

```

```

RETURN
END

```

```

DATA SET REDUCE AT LEVEL 007 AS OF 83/02/24
SUBROUTINE REDUC(A,VV,MAXA,NEQB,NWA,NFQ,NBLOCK,M1,MA,NCALL,KKS) 00001**4
IMPLICIT REAL*8(A-H,O-Z) 00002
00003
CALLED BY: SUBSTP 00004
00005
THIS ROUTINE REDUCES AND BACK-SUBSTITUTES A SINGLE VECTOR STORED 00006
IN CORE USING A REDUCED MATRIX STORED IN BLOCK FORM. 00007
00008
DIMENSION A(NWA),VV(NFQ),MAXA(M1) 00009
00010
COMMON/TAPED/NDSTP,NRED,NB,NR,IFIL(2),NUMBER 00011**7
00012
COMMON/GORA/IFLAG,ITR 00013**6
INC=NEQB - 1 00014
***** CHECK PASSING ARGUMENT ITR = 0 FOR ITERATION ***** 00015**5
00016**5
MA1 = MA - 1 00017
00018
PERFORM FORWARD REDUCTION OF THE VECTOR 00019
GO TO (10,10,15,10),KKS 00020**3
00021**3
00022
10 IF (NBLOCK.EQ.1.AND.NCALL.GT.1) GO TO 22 00023**3
GO TO 30 00024**3
15 CONTINUE 00025**3
00026**5
00027**5
IF (IPR.NE.0) GO TO 22 00028**5
00029**5
30 REWIND NRED 00030**3
00031**3
READ (NRED) A,MAXA 00032
22 ISA = 1 00033
KSTART = 2 00034
KEND = NRED 00035
00036
500 N = 1 00037
DO 100 K=KSTART,KEND 00038
N = N+1 00039
KL=N + INC 00040
KU=MAXA(N) 00041
IF (KU-KL) 100,110,110 00042
110 KU = K 00043
DO 130 KK=KL,KU,INC 00044
KJ=KU - 1 00045
130 VV(K)=VV(K) - A(KK)*VV(KJ) 00046
100 CONTINUE 00047
00048
IF (ISA.EQ.NBLOCK) GO TO 175 00049
KL = NEQB 00050
ML = KEND+1 00051
MR = MIN0(KEND+MA1,NEQB) 00052
N = NEQB 00053
DO 140 K=ML,MR 00054
N = N+1 00055
KL=KL + NEQB 00056
KU=MAXA(N) 00057
IF (KU-KL) 140,150,150 00058
150 KJ = KEND 00059

```

	DO 170 KK=KI, KU, INC	00060
	VV(K) = VV(K) - A(KK)*VV(KJ)	00061
170	KJ=KJ - 1	00062
140	CONTINUE	00063
		00064
175	KSI = KSTART-1	00065
	N = 0	00066
	DO 200 K=KSI, KEND	00067
	N = N+1	00068
	C = A(N)	00069
	IF (C) 180, 200, 180	00070
180	VV(K) = VV(K)/C	00071
200	CONTINUE	00072
205	IF (ISA.EQ.NBLOCK) GO TO 400	00073
	READ (NREQ) A, MAXA	00074
	ISA=ISA+1	00075
	KSTART = KSTART+NEQB	00076
	KEND = MINO(KEND+NEQB, NEQ)	00077
		00078
	GO TO 500	00079
		00080
	BACK-SUBSTITUTE REDUCED VECTOR (STORED IN CORE)	00081
		00082
400	IF (ISA.GT.1)	00083
	*BACKSPACE NREQ	00084
	ISA=1	00085
	NW = NEQ-(NBLOCK-1)*NEQB	00086
	KEND = NEQ	00087
	GO TO 645	00088
		00089
420	KEND = KEND-NW	00090
	NW = NEQB	00091
		00092
	KI=NEQB	00093
	MK = MINO(NEQ, KEND+MA1)	00094
	ML = KEND+1	00095
	N = NEQB	00096
	DO 600 K=KI, MK	00097
	N = N+1	00098
	KI=KI+NEQB	00099
	KU=MAXA(N)	00100
	IF (KU-KI) 600, 610, 610	00101
610	KJ = KEND	00102
	DO 620 KK=KI, KU, INC	00103
	VV(KJ)=VV(KJ) - A(KK)*VV(K)	00104
620	KJ=KJ - 1	00105
600	CONTINUE	00106
		00107
645	N = NW	00108
	K = KEND	00109
	DO 640 I=2, NW	00110
	KI=N + INC	00111
	KU=MAXA(N)	00112
	IF (KU-KI) 655, 650, 650	00113
650	KJ=K	00114
	DO 690 KK=KI, KU, INC	00115
	KJ=KJ - 1	00116
690	VV(KJ)=VV(KJ) - A(KK)*VV(K)	00117
655	N=N - 1	00118
640	K = K-1	00119

DATA SET SDFDMA AT LEVEL 031 AS OF 83/03/05
 SUBROUTINE SDFDMA(VOL,EM,IMAT,KS,KE,NEI,THICK)

IMPLICIT REAL*8(A-H,O-Z)

COMMON/CONTI/DT,HEP(8),RADH,NASIM,NUMEL,NEQ,MBAND,NCYCL,NDYN,IPLNA
 *X,ISTPRK,NPKRST,ISTART,NEGH,MBANDH,NEGB,MBANDS
 COMMON/EMB/ S(16,16),GM(16),GH(16,16),SS(16,16),XP(16)
 *,AN(16),LM(16)
 COMMON/QUAD/RR(4),ZZ(4),FAC,R,Q(6),P(4,12),SF(12,12),H(12,12),
 *SI(20,12),b(4,4),C(12,12),E(12,12),TT(5,12),PI(4,8)
 COMMON/BELA/NS,NEI,LYF,ND,NUMGT,MODEL,NCUN,LDW,NKU15,NKU16,NKU17
 COMMON/BOBBY/KKS,NUMMAT,NG,LOOP,NORN,NINT,LDWA,NPT
 COMMON/ASIK/C1,CZ,C3,AL,ALF,FACC,FACT

DIMENSION IN(8),LN(6),EM(1)

***** CALCULATE FLUID STIFFNESS MATRIX (EVALUATE AT CE

K=LMAT
 FAC=VIM.*H
 FACM=FAC*EM(K)
 DO 505 I=1,8
 DO 505 J=1,8
 E(I,J)=E(I,J)+PI(1,1)*TT(1,J)+FACM
 C(I,J)=C(I,J)+E(I,J)*FACC
 SF(I,J)=SF(I,J)+E(I,J)*ALF

505 CONTINUE
 952 CONTINUE

IF(NG.GT.0.AND.KKS.EQ.3) GO TO 6200
 IF(NEI.NE.0) GO TO 6200

PRINT 999
 999 FORMAT(//)

WRITE(6,4)
 4 FORMAT(1H1,' ***** UNDRAINED ANALYSIS ***** '/')
 WRITE(6,225) ((SF(I,J),J=1,8),I=1,8)
 PRINT 999

WRITE(6,66)
 66 FORMAT(5X,'COUPLING FLUID-SOLID MATRIX')
 WRITE(6,225) ((C(I,J),J=1,8),I=1,8)
 PRINT 999
 WRITE(6,69)

```

69  FORMAT(5X,'FLUID MATRIX')
WRITE(6,225) ((C(I,J),J=1,8),I=1,8)
225  FORMAT(10X,'#12.5)
      PRINT *,*
72  FORMAT(5X,'FLUID DISSIPATION MATRIX')
WRITE(6,225) ((H(I,J),J=1,8),I=1,8)
6200 CONTINUE
*****
600 CONTINUE
WRITE(51) ((SF(I,J),J=1,8),I=1,8),((E(I,J),J=1,8),I=1,8),((C(I,J),J=1,8),I=1,8),
*,J=1,8),I=1,8),IMAT,KS,KF,ALF,((H(I,J),J=1,8),I=1,8)
*,(XP(I),I=1,16), (AN(I),I=1,16)
500 CONTINUE
RETURN
END
00060
00061
00062
00063
00064
00065
00066
00067
00068
00069
00070
00071
00072
00073*23
00074*23
00075
00076*6
00077*8
00078*28
00079*28
00080*28
00081*15
00082
00083

```

```

DATA SET SECND AT LEVEL 001 AS OF 81/07/03
SUBROUTINE SECND (A,B,V,MAXA,N,VV,ROOT,TIM,ERRVL,ERRVR,
INTE,N,NA,NROUT,NC,IFPR,ANORM,CUFO)
IMPLICIT REAL*8(A-N,U-Z)
REAL*4 TIM1,TIM2,TIM3
CALLS: BANDET
CALLED BY: MUDS
COMMON /TAPES/NSTIF,NRED,WL,NR,NT,NMASS
*****CHANGED *****K*****
DIMENSION A(N,NA),B(N),V(1),W(1),VV(N,1),WW(N,1),ROOT(1),
1TIM(1),ERRVL(1),ERRVR(1)
*****CHANGED *****R*****
INTEGER NITE(1),MAXA(1)
COMMON /EM/ AT(1000),IFIBD(3138)
COMMON /ASPHLD/NM01K
THE FOLLOWING TOLERANCES ARE SET FOR THE IBM 370
ACTOL=1.0D-04
RCBTOL=1.0D-05
RTOL=1.0D-10
RUTOL=1.0D-12
** SCALE=2.000**ZOO
SCALE = 1.7000D+38
NPF=5
1ITEM=10
NITEM=00
NVM=0
REWIND 01
REWIND NMASS
READ (NMASS) B
PIA = 2.0
NOV=0
JR=1
NSA=0
NWA=N+NA
LSC=1000
FIND LOCATIONS FOR NEGATIVE ELEMENTS IN STARTING ITERATION VECTORS
REWIND NSTIF
READ (NSTIF) (A(1,I),I=1,N)
DO 1 I=1,N
AA=A(1,I)
IF (AA.GT.0.) GO TO 1
WRITE (0,1000) I,AA
STOP
1 V(1)=B(1)/AA
DO 2 J=3,NC
RMAX=0.
DO 3 I=1,N
IF (V(1)-V(I).GT.RMAX) GO TO 3
RMAX=V(1)
IMAX=I
3 CONTINUE
NITE(0)=IMAX

```

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00001
00002
00003
00004
00005
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00010
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00057
00058

```

2	V(LMAX)=0.	00059
	CHECK FOR SINGLE DEGREE-OF-FREEDOM SYSTEM	00060
		00061
	IF (N.GT.1) GO TO 5	00062
	IF (B(1).GT.0.) GO TO 7	00063
	WRITE(6,1180)	00064
	STOP	00065
7	REWIND NSTIF	00066
	READ(NSTIF) A(1,1)	00067
	ROOT(1)=A(1,1)/B(1)	00068
	NSCH=1	00069
	A(1,1)=1.000/DSORT(B(1))	00070
	GO TO 950	00071
		00072
		00073
5	CONTINUE	00074
	RA=0.0	00075
	RR=0.0	00076
	CALL BANDET (A,B,V,MAXA,N,NWA,RA,NSCH,DETA,ISC,1)	00077
	FA=DETA	00078
	FR=FA	00079
	DETR=DETA	00080
		00081
	CHECK FOR ZERO EIGENVALUE(S)	00082
	WRITE(6,1002) ANORM	00083
1002	FORMAT(5X,F12.5)	00084
	IF (A(N,1) .GT. ANORM) GO TO 10	00085
	WRITE (6,1003)	00086
	STOP	00087
		00088
	FIND LOWER BOUND ON SMALLEST EIGENVALUE	00089
10	IF (IFPR.EQ.1)	00090
*	WRITE(6,1010)	00091
	DO 100 I=1,N	00092
100	W(1)=B(1)	00093
	KT=0.6	00094
	ITTE=0	00095
	KK=2	00096
110	ITTE=ITTE+1	00097
	DO 120 I=1,N	00098
120	V(I)=W(1)	00099
	CALL BANDET (A,B,V,MAXA,N,NWA,RA,NSCH,DETA,ISC,KK)	00100
	KK=2	00101
	RQT=0.0	00102
	DO 130 I=1,N	00103
130	RQI=RQI+W(I)+V(I)	00104
	DO 180 I=1,N	00105
180	W(I)=R(I)*V(I)	00106
	RQB=0.0	00107
	DO 140 I=1,N	00108
140	RQB=RQB+W(I)+V(I)	00109
	RG=RQI/RQB	00110
	IF (IFPR.EQ.1)	00111
*	WRITE (6,1004) RG	00112
	BS=DSORT(RQB)	00113
	TOL=DATA5(KG-KI)/RG	00114
	IF (TOL.LT.RCBTOL) GO TO 150	00115
	DO 160 I=1,N	00116
160	W(1)=W(1)/BS	00117
	RT=RG	00118

	IF (111E.L1.111E) GO TO 110	00119
150	DO 170 I=1,N	00120
170	V(I)=V(I)/RS	00121
	RB=RQ*(1.000-DMIN1(1.00-1,1.002*TOL))	00122
	IS=0	00123
230	CALL BANDET (A,B,V,MAXA,N,NWA,RB,NSCH,DETB,ISC,1)	00124
	IF (11PR.EQ.1)	00125
*	WRITE (6,1020) RB,NSCH	00126
	FB=DETB	00127
	IF (NSCH.EQ.0) GO TO 300	00128
	IS=IS+1	00129
	IF (IS.LE.NTF) GO TO 240	00130
	WRITE (6,1030) NTF	00131
	STOP	00132
240	RB=RB/(NSCH+1)	00133
	GO TO 230	00134
		00135
	ITERM FOR INDIVIDUAL ROOTS	00136
300	IF (11PR.EQ.1)	00137
*	WRITE (6,1040)	00138
	NITE(JK)=-1	00139
	IF (11PR.EQ.1)	00140
*	WRITE (6,1050) JK,NITE(JK),RA,DETA,FA,ETA,ISC	00141
	NITE(JK)=0	00142
	IF (11PR.EQ.1)	00143
*	WRITE (6,1050) JK,NITE(JK),RB,DETB,FB,ETA,ISC	00144
		00145
	WE STOP WHEN WE HAVE THE REQUIRED NO OF ROOTS SMALLER THAN RC AND	00146
	NOV=0	00147
310	IF (NSCH.GE.NROD1) GO TO 900	00148
	IF (RB.GT.CHI0) GO TO 900	00149
		00150
03	DIF=FB-FA	00151
	IF (DIF.NE.0.0) GO TO 320	00152
	WRITE (6,1060)	00153
	GO TO 900	00154
320	DEL=FB*(RB-RA)/DIF	00155
	RC=RB-ETA*DEL	00156
	TOL=RCB100+RC	00157
	IF (DABS(RC-RB) .GT. TOL) GO TO 330	00158
	IF (11PR.EQ.1)	00159
*	WRITE (6,1070)	00160
	ROOT(JK)=RB	00161
	GO TO 400	00162
		00163
		00164
330	CALL BANDET (A,B,V,MAXA,N,NWA,RC,NSCH,DETC,ISC,1)	00165
	FC=DETC	00166
	NITE(JK)=NITE(JK)+1	00167
	IF (JK.EQ.1) GO TO 340	00168
	JJ=JK-1	00169
	DO 350 K=1,00	00170
350	FC=FC/(RC-ROOT(K))	00171
340	IF (11PR.EQ.1)	00172
*	WRITE (6,1050) JK,NITE(JK),RC,DETC,FC,ETA,ISC	00173
		00174
	IF WE HAVE MORE SIGNCHANGES THAN EIGENVALUES SMALLER THAN RC WE	00175
	START INV. ITERATION	00176
	NES=0	00177
	IF (JK.EQ.1) GO TO 300	00178

	DU 360 I=1,50	00179
360	IF (RDOT(1).LT.RC) NES=NFS+1	00180
380	NOV=NSCH-MS	00181
	IF (NOV.EQ.0) GO TO 370	00182
	IF (IFPR.EQ.1)	00183
*	WRITE (6,1080) NOV	00184
	RDOT(JK)=RC	00185
	IF (NOV.GT.1) NSK=1	00186
		00187
	GO TO 400	00188
370	RR=RA	00189
	FR=FA	00190
	DETR=DETA	00191
	RA=RB	00192
	FA=FB	00193
	DETA=DETB	00194
	RB=RC	00195
	FB=FC	00196
	DETB=DETC	00197
		00198
	WE RESET ETA IF NECESSARY	00199
	TOL=RR*AC101	00200
	IF (DABS(RA-RB).LT. TOL) ETA=ETA*2.000	00201
	IF (NITE(JK).GE.NITEM) GO TO 310	00202
	WRITE (6,1015) JK,NITE(JK)	00203
	GO TO 900	00204
		00205
	CHECK FOR STORAGE	00206
400	IF (JK.LE.NC) GO TO 405	00207
	WRITE (6,1090)	00208
	GO TO 900	00209
		00210
405	NUR=JK-1	00211
	IF (NUR.GT.NVM) NUR=NVM	00212
	IF (IFPR.EQ.1)	00213
*	WRITE (6,1100) NUR	00214
	IF (JK.EQ.1) GO TO 410	00215
	DU 420 I=1,N	00216
420	V(I)=1.0	00217
	KK=2	00218
	IF (JK.EQ.NC) GO TO 410	00219
	I = NITE(JK+1)	00220
	V(I) = -1.	00221
410	DU 430 I=1,N	00222
430	W(I)=B(I)*V(I)	00223
	IS=0	00224
	GO TO 510	00225
		00226
	INVERSE 11PRN	00227
440	NITE(JK)=NITE(JK)+1	00228
	DU 450 I=1,N	00229
450	V(I)=W(I)	00230
	CALL HANDP1 (A,B,V,MAXA,N,NWA,RC,NSCH,DETC,ISC,KK)	00231
	IF (IS.EQ.1) GO TO 460	00232
	KK=2	00233
	KQT=6.0	00234
	DU 470 I=1,N	00235
470	KQT=KQT+W(I)*V(I)	00236
	DU 475 I=1,N	00237
475	W(I)=B(I)*V(I)	00238

	KQB=0.0	00239
480	DO 480 I=1,N	00240
	KQB=KQB+W(I)*V(I)	00241
	KQ=KQT/RQB	00242
	KT=ROOT(JK)+RQ	00243
	IF (IFPK.EQ.1)	00244
*	WRITE (6,1110) JK,NITE(JK),KT,RQ	00245
	TQB=KT+KQTQB	00246
	IF (DAHS(KT-R1A) .GT. TQB) GO TO 510	00247
	IS=1	00248
	GO TO 440	00249
		00250
510	KTA=KT	00251
	BS=DSQR(KQB)	00252
	DO 490 I=1,N	00253
490	W(I)=W(I)/BS	00254
	IF (NOR.EQ.0) GO TO 550	00255
	DO 520 K=1,NOR	00256
	AL=0.0	00257
	DO 530 I=1,N	00258
530	AL=AL+V(I,K)*W(I)	00259
	DO 540 I=1,N	00260
540	W(I)=W(I)-AL*W(I,K)	00261
520	CONTINUE	00262
		00263
550	IF (NITE(JK).LE.NITEM) GO TO 440	00264
	WRITE (6,1015) JK,NITE(JK)	00265
	GO TO 500	00266
		00267
460	KQT=0.0	00268
	ERRT=KQB	00269
	DO 570 I=1,N	00270
570	KQT=KQT+V(I)*W(I)	00271
	DO 560 I=1,N	00272
560	W(I)=W(I)*V(I)	00273
	KQB=0.0	00274
	DO 580 I=1,N	00275
580	KQB=KQB+V(I)*W(I)	00276
		00277
		00278
	OBTAI N A RATHER LARGE ERROR BOUND	
	KQ=KQT/RQB	00279
	KOUT(JK)=ROOT(JK)+RQ	00280
	ERR=DSQR(T(ERRT/RQB))	00281
	ERRVD(JK)=KOUT(JK)-ERR	00282
	ERRVK(JK)=KOUT(JK)+ERR	00283
		00284
		00285
	BS=DSQR(KQB)	00286
	DO 590 I=1,N	00287
590	W(I)=W(I)/BS	00288
	V(I)=V(I)/BS	00288
	JJ=JK	00289
	IF (DD.LE.NVM) GO TO 610	00290
	WRITE (N1) (VV(J,I),J=1,N)	00291
	DO 600 K=1,N	00292
	DO 600 L=2,NVM	00293
	WW(K,L-1)=WW(K,L)	00294
600	VV(K,L-1)=VV(K,L)	00295
	JJ=NVM	00296
610	DO 620 K=1,N	00297
	WW(K,JJ)=V(K)	00298

620	VV(K, JJ)=V(K)	00299
		00300
		00301
	DECIDE STRATEGY FOR ITERM, TOWARDS NEXT ROOT	00302
	TOL=RTOL*ROOT(JK)	00303
	IF (NOV.GT.0) GO TO 700	00304
	IF (DABS(ROOT(JK)-RB) .GT. TOL) GO TO 710	00305
	IF (RA.GT.0.0) GO TO 720	00306
	RA=RB/2.	00307
	CALL BANDET (A,B,V,MAXA,N,NWA,RA,NSCH,DETA,ISC,1)	00308
	FA=DETA	00309
720	RB=RA	00310
	FB=FA	00311
	DETB=DETA	00312
	RA=RK	00313
	FA=FK	00314
	DETA=DETR	00315
	GO TO 710	00316
		00317
700	IF (ROOT(JK).GT.RC) NSK=1	00318
	IF (NSK.EQ.1) GO TO 730	00319
	IF (DABS(RC-ROOT(JK)) .LT. TOL) GO TO 740	00320
	IF (DABS(ROOT(JK)-RB) .LT. TOL) GO TO 750	00321
	RA=RB	00322
	FA=FB	00323
	DETA=DETB	00324
750	RB=RC	00325
	FB=FC	00326
	DETB=DETC	00327
	GO TO 710	00328
740	IF (DABS(ROOT(JK)-RB) .GT. TOL) GO TO 710	00329
	IF (RA.GT.0.0) GO TO 760	00330
	RA=RB/2.	00331
	CALL BANDET (A,B,V,MAXA,N,NWA,RA,NSCH,DETA,ISC,1)	00332
	FA=DETA	00333
760	RB=RA	00334
	FB=FA	00335
	DETB=DETA	00336
	RA=RK	00337
	FA=FK	00338
	DETA=DETR	00339
710	FA=FA/(RA-ROOT(JK))	00340
	FB=FB/(RB-ROOT(JK))	00341
	JR=JK+1	00342
	ETA=2.0	00343
	GO TO 300	00344
		00345
730	IF (RA.GT.0.0) GO TO 780	00346
	RA=RB/2.	00347
	CALL BANDET (A,B,V,MAXA,N,NWA,RA,NSCH,DETA,ISC,1)	00348
	FA=DETA	00349
780	IF (DABS(ROOT(JK)-RB).GT.TOL) GO TO 770	00350
	RB=RA	00351
	FB=FA	00352
	DETB=DETA	00353
	RA=RK	00354
	FA=FK	00355
	DETA=DETR	00356
770	FA=FA/(RA-ROOT(JK))	00357
	FB=FB/(RB-ROOT(JK))	00358

	FR=FR/(FR-ROOT(JK))	00359
	IF (ROOT(JK).LE.RC) NOV=NOV-1	00360
	JR=JR+1	00361
	NITE(JK)=0	00362
	ROOT(JK)=RC	00363
	IF (NOV.GI.0) GO TO 400	00364
	NSK=0	00365
	ETA=2.0	00366
	GO TO 300	00367
900	NROOT=JK-1	00368
	IF (NROOT.GI.0) GO TO 902	00369
	WRITE (6,1180)	00370
	STOP	00371
902	CONTINUE	00372
	IF (1+FR.EQ.0) GO TO 905	00373
	WRITE (6,1140)	00374
	WRITE (6,1006) (NITE(J),J=1,NROOT)	00375
	WRITE (6,1150)	00376
	WRITE (6,1008) (TIM(J),J=1,NROOT)	00377
	WRITE (6,1160)	00378
	WRITE (6,1004) (ERRVL(J),J=1,NROOT)	00379
	WRITE (6,1004) (ERRVK(J),J=1,NROOT)	00380
		00381
		00382
	READ EIGENVECTORS INTO CORE	00383
905	IF (NROOT.LE.NVM) GO TO 906	00384
	NDIF=NROOT - NVM	00385
	REWIND NI	00386
	DO 904 L=1,NDIF	00387
	READ (NI) (A(1,L),I=1,N)	00388
904	CONTINUE	00389
	GO TO 908	00390
906	NDIF=0	00391
908	NROOT=NROOT - NDIF	00392
	DO 912 L=1,NROOT	00393
	DO 912 I=1,N	00394
912	A(1,L+NDIF)=VV(1,I)	00395
		00396
	ARRANGE EIGENVALUES AND VECTORS IN ASCENDING ORDER	00397
	IF (JK.LE.2) GO TO 950	00398
	JR=JK-2	00399
910	IS=0	00400
	DO 920 I=1,JR	00401
	IF (ROOT(I+1).GE.ROOT(I)) GO TO 920	00402
	IS=IS+1	00403
	KT=ROOT(I+1)	00404
	ROOT(I+1)=ROOT(I)	00405
	ROOT(I)=KT	00406
	DO 930 K=1,N	00407
	KT=A(K,I+1)	00408
	A(K,I+1)=A(K,I)	00409
930	A(K,I)=KT	00410
920	CONTINUE	00411
	IF (IS.GI.0) GO TO 910	00412
		00413
950	WRITE (6,1170)	00414
	NROOT=NSCH	00415
	WRITE (6,1004) (ROOT(J),J=1,NROOT)	00416
		00417
	CALCULATE PHYSICAL ERROR NORMS	00418

	REWIND N1	00419
	DO 955 I=1, NR001	00420
955	WRITE (N1) (A(K,L), K=1, N)	00421
	REWIND NS11F	00422
	READ (NS11F) (A(1,I), I=1, NWA)	00423
	REWIND NT	00424
	DO 960 I=1, NR001	00425
	RT = ROOT(L)	00426
	READ (N1) (V(1), I=1, N)	00427
	CALL MULT (W, A, V, N, MA)	00428
	VNORM=0.	00429
	DO 958 I=1, N	00430
958	VNORM = VNORM + W(1)*W(1)	00431
	DO 960 I=1, N	00432
960	W(I) = W(I) - RT*A(I)+V(I)	00433
	WNORM = 0.0	00434
	DO 968 I=1, N	00435
968	WNORM = WNORM + W(1)*W(1)	00436
	VNORM = DSQRT(VNORM)	00437
	WNORM = DSQRT(WNORM)	00438
	ERRVL(L) = WNORM/VNORM	00439
960	CONTINUE	00440
	REWIND N1	00441
	DO 969 I=1, NR001	00442
969	READ (N1) (A(K,L), K=1, N)	00443
	WRITE (6, 1190)	00444
	WRITE (6, 1004) (ERRVL(J), J=1, NR001)	00445
	REWIND N1	00446
	DO 970 I=1, NR001	00447
970	ROOT(I) = DSQRT(ROOT(I))	00448
	WRITE (N1) (ROOT(I), I=1, NR001)	00449
	NWA=N*NR001	00450
	WRITE (N1) (A(1,I), I=1, NWA)	00451
	P12=8.000*ATAN(1.000)	00452
	DO 980 I=1, NR001	00453
980	AT(I) = P12/ROOT(I)	00454
	RETURN	00455
		00456
		00457
		00458
		00459
1000	FORMAT (4H ***ERROR NEG OR ZERO DIAGONAL ELEMENT A(,14,4H) = ,	00460
1	11H, 21H BEFORE DECOMPOSITION)	00461
1004	FORMAT (1H0, 6E20.12)	00462
1006	FORMAT (1H0, 6I20)	00463
1008	FORMAT (1H0, 6E20.2)	00464
1009	FORMAT (43H0***ERROR SOLUTION TERMINATED IN *SECANT*, /	00465
1	12X, 25H RIGID BODY MODE(S) FOUND., / 1X)	00466
1010	FORMAT (51H1 INVERSE ITERATION GIVES FOLLOWING APPROXIMATION TO,	00467
1	18H LOWEST EIGENVALUE, 1X)	00468
1015	FORMAT (41H0***ERROR THE *FACTOR EXIT FROM *SECANT*, / 12X,	00469
1	37H ITERATION ABANDONED FOR ROOT NUMBER = , 14 / 12X,	00470
2	37H NUMBER OF ITERATIONS PERFORMED = , 14 / 1X)	00471
1020	FORMAT (5H0R = E20.12, 7H NSCH = 14)	00472
1030	FORMAT (30H0***ERROR SOLUTION STOP IN *SECANT*, / 12X, 1H(00473
1	13, 48H) FACTORIZATIONS PERFORMED IN AN ATTEMPT TO FIND,	00474
2	32H LOWER BOUND ON FIRST EIGENVALUE, / 12X,	00475
3	16H CHECK THE MODEL., / 1X)	00476
1040	FORMAT (1H1, 4X, 4H ROOT, 4X, 4H TIME, 10X, 2H RC, 15X, 12H DEL (A-RC*B), 15X,	00477
		00478

	/2HFC,13A,3HFTA,4X,3HISC)	00479
1050	FORMAT (1H0,4X,14,4X,14,8X,3F22.14,F/.2,16)	00480
1060	FORMAT (42H0THE DEFLATED POLYNOMIAL HAS NO MORE ROOTS)	00481
1070	FORMAT (29H0(C-C-RO) IS SMALLER THAN 100)	00482
1080	FORMAT (16H0WE JUMPED OVER 14,16H UNKNOWN ROOT(S))	00483
1090	FORMAT (41H0**ERROR PRE-MATURE EXIT FROM *SECND*,	00484
1	34H CAUSED BY EITHER OF THE FOLLOWING, / 12X,	00485
2	22H(1) BAD MODEL DATA, OR, / 12X,	00486
3	52H(2) ROOT CLUSTER (I.E., NEAR EQUAL OR REPEATED EIGEN,	00487
4	30HVALUES) ENCOUNTERED AT CURRENT SHIFT, / 16X,	00488
5	25HCAUSING STORAGE OVER-FLOW, 1X)	00489
1100	FORMAT (1H0,34X,4HROOT,16X,2HR0,16X,4HNOK=,12)	00490
1110	FORMAT (1H0,4X,14,4X,14,8X,2E22.14)	00491
1140	FORMAT (42H0NO OF ITERATIONS FOR EACH EIGENVALUE ARE /)	00492
1150	FORMAT (30H0TIME USED FOR EACH EIGENVALUE /)	00493
1160	FORMAT (43H0FOLLOWING ARE ERROR BOUNDS ON EIGENVALUES)	00494
1170	FORMAT (/// 40H WE SOLVED FOR THE FOLLOWING EIGENVALUES)	00495
1180	FORMAT (37H0**ERROR SOLUTION STOP IN *SECND*, / 12X,	00496
1	23HNO EIGENVALUES COMPUTED, / 1X)	00497
1190	FORMAT (/// 40H THE FOLLOWING ARE PHYSICAL ERROR BOUNDS,	00498
1	20H ON THE EIGENPAIRS)	00499
	END	00500
		00501

```

DATA SET SHAKE AT LEVEL 059 AS OF 83/02/11
SUBROUTINE SHAKE(NPK,NPJ,KP1,KK,IQ,IH,AM,EK,AN,ANI,N43,A1,NPT,
*GB,SO)
IMPLICIT REAL*8 (A-H,O-Z)
COMMON D(1)
COMMON/CONTI/DO,HED(8),KADN,NUMNP,NOMED,NEQ,MBAND,NCYCL,NBYN,IPLNA
*X,ISTRPK,NPRINT,ISTARI,NEQH,MBANDH,NEQS,MBANDS
COMMON/ASIM/IIP,NSC,NSLC,NLC,NNYP,NNSC
COMMON/IARA/LPC,N,M1,M2,NGT,NGF,NPRINT,M,MM
COMMON/DIMBL/M43,IPT,IPT1,MK,MPRINT
COMMON/GUAI/CCCC(1500),N1,N2,N3,N4,N5,MTOT,NNN,NK15
DIMENSION ID(NUMNP,1),IH(NUMNP,1),AM(1),EK(1),AN(NPJ,1),ANI(NYP,1)
*,AI(NEQS,1)
*,GB(1),SU(1)
N44=N43+NPJ+NPJ
N45=N44+NPJ*NEQS
N045=N45+NPJ*NEQS
N046=N045+NPJ*NPJ
IF(IPT.EQ.1.AND.IPT1.EQ.1.OR.IPT.EQ.1.AND.IPT1.EQ.2)N046=N045+NPJ
NGT=NPT
NGF=KP1
NPRINT=MPRINT
IF(NPT.EQ.0) GO TO 3
1 CONTINUE
2 CONTINUE
***** CALL SHAKEM AT THIS STAGE *****
3 CONTINUE
N047=N046+NEQS*NEQS
N050=N047+NEQS
N051=N050+NEQS
N052=N051+NEQ

```

```

00001*41
00002*41
00003*52
00004*41
00005*41
00006
00007*20
00008**7
00009**7
00010**7
00011*10
00012**7
00013**7
00014*54
00015**6
00016*55
00017*55
00018*39
00019**6
00020**6
00021*29
00022*29
00023*14
00024*52
00025**6
00026*39
00027*39
00028*39
00029**6
00030*26
00031*26
00032*40
00033*22
00034*34
00035*42
00036*55
00037*11
00038*11
00039**6
00040**6
00041*58
00042**6
00044
00046**6
00047**6
00048**6
00049
00050
00051*43
00052*43
00053*43
00054*11
00055*11
00056
00057*11
00058*24
00059*26
00060*21

```



```

N053=N052+NPJ
N054=N053+NPJ+NPJ
IF (IPT.EQ.1.AND.(IPT1.EQ.1.OR.IPT.EQ.1.AND.IPT1.EQ.2))N054=N053+NPJ

```

00061*21

```
IF(KPT.EQ.1) GO TO 4
```

```
***** ALLOCATION FOR ZX3LP = MAXIMIZATION *****
```

```

MN=NPJ+1+Z
IF (NPT.EQ.1) MN=(NPT+NEQS)+1+Z
IF (NPT.EQ.2) MN=(NPJ+NEQS)+1+Z

N055=N054+MN

M1=NPJ
M2=0
IF (NPT.EQ.1) M2=NEQS
IF (NPT.EQ.2) M2=NEQS
LPC=(M1+M2+2)+1

MM=M1+M2+1
N=NPJ+1

IF (NPT.EQ.1.OR.NPT.EQ.2) N=(1+Z*NPJ)

N050=N055+LPC*N
N052=N050+N+2
N053=N052+MAX0(N,M1+M2+1)
N054=N053+LPC
N055=N054+LPC+LPC+3*M1+Z*M2+4
N056=N055+2*M2+3*M1+4

***** DUMMY DIMENSION FOR CUPT,ICULMS,IDES,&ROW *****

```

00062*28

00063*33

00064*33

00065*43

00066*28

00067*28

00069*29

00070*27

00071*27

00072*27

00073*27

00074*27

00075*33

00076*43

00077*28

00078*28

00079*28

00080*27

00081*33

00082*33

00083*45

00084*43

00085*27

00086*33

00087*48

00088*27

00089*27

00090*27

00091*48

```

NH50=N050+(NPJ+NPJ)
N057=NH50+1
N058=N057+1
N059=N058+1

```

```
GO TO 6
```

```
***** ALLOCATION FOR ZXMIN= MINIMIZATION *****
```

```
4 CONTINUE.
```

```

M1=0
M2=NPJ+1
IF (NPT.EQ.1.OR.NPT.EQ.2) M2=NPJ+1
LPC=M2+Z

```

M=MZ+1
MM=M+1
N=NPK+2+NRUD

NU55=NU54+(MM)
NU56=NU55+(MM*N)
N52=NU56+N+1
N53=N52+MM
N54=N53+MM
N55=N54+N
N56=N55+1
NM56=N56+(MM+MM)
NU57=NM56+LPC
NU58=NU57+LPC
NU59=NU58+LPC

***** FINISH *****

0 CONTINUE

CHECK FOR BOUND *****

N57=NU59+MAXU(N,M1+MZ+1)

IF (KK.EQ.0) N57=NU59+1
N58=N57+LPC

IF (KK.EQ.0) N58=N57+1
N59=N58+NPK

IF (KK.EQ.0) N59=N58+1

N60=N59+NPK+1

IF (KK.EQ.0) N60=N59+NPK+1

WRITE (0,106)N43,N44,N45,N045

WRITE (0,108)N046,N047,N050,N051,N052,N053,N054,N055

WRITE (0,108)N056,N52,N53,N54,N55,N56,NM56,N057,N058

WRITE (0,108)N059,N57,N58,N59,N60

IF (N60.GT.M101) GO TO 900

CALL ACUFP(D(N43),D(N44),D(N45),NUMEL,NPJ,NPK,AN,NPT)

IF (NPT.EQ.0) GO TO 800

IF (NPT.EQ.2.OR.NPT.EQ.1) GO TO 110

CALL STIFF(A1,NEWS,MPANDS,D(N046))

00092*48

00093*48

00094*48

00095*56

00096*31

00097*56

00098*56

00099*31

00100*11

00101*56

00102*52

00103*56

00105*25

00104*11

00105*25

00106*37

00107*52

00108*52

00109*52

00110*49

00111*49

00112*49

00113*49

00114*49

00115*57

00117*57

00118*57

00119*57

00120*54

00121*58

00122*54

00123*54

00124*49

00125*11

00126*24

00127*54

00128*54

00129*54

110 CONTINUE

00130*11
00131*11
00132*17
00133*17
00134*49

108 FORMAL(' ',10112)
MU=MPJ
MU=MPK
PRINT 109
PRINT 109

9 FORMAT(' *** RETURN FOR SHAKE *****')

```

*****
N43 = S
N44 = C
N45 = (T
N46 = AWIT
N47 = AS
N48 = SI
N49 = SM
N50 = SW
N51 = SP
N52 = SD
N53 = SW
N54 = A
N55 = CB
N56 = PSUL
N57 = DSUL
N58 = KW
N59 = IW

```

```

CALL INPTX(D(N43),D(N44),D(N45),AR,D(N46),D(N47),D(N48),D(N49),D(N50),
D(N51),D(N52),D(N53),D(N54),D(N55),D(N56),D(N57),D(N58),D(N59)),
D(N52),D(N53),D(N54),D(N55),AM,EK,KK,IM,D(N56),D(N57),D(N58),D(N59))

```

GO TO 900
800 CONTINUE

```

***** ALLOCATION FOR SHAKE *****
00135*17
00136*11
00137*11
00138*29
00139*29
00140*29
00141*29
00142*29
00143*29
00144*29
00145*29
00146*29
00147*29
00148*29
00149*29
00150*29
00151*29
00152*29
00153*29
00154*29
00155*29
00156*29
00157*29
00158*11
00159*24
00160*29
00161*29
00162*27
00163*29
00164*29
00165*41
00166*41
00167*52
00168*11
00169*11
00170*11
00171*21
00172*22
00173
00174
00175
00176*6
00177*6
00178*27
00179*11
00180*11
00181*11
00182
00183
00184
00185

```

900 CONTINUE
RETURN
END

```

DATA SET S01          AT LEVEL 049 AS OF 83/02/24
SUBROUTINE SUD(A),XMS,NEQS,MBANDS,NUMNP,MBETA,KKS,ID,IH,LOOP,NDUM,00001*11
*DBT1,DBT2,DBETA1,GMX,DPL,GUSE,ES,EN,PU,AS,X,ROOT,VROOT,KOBT1) 00002*31
IMPLICIT REAL*8(A-H,O-Z) 00003*25
COMMON D(1) 00004*31
00005*19
00006*20
00007*31
00008**6
00009**6
00010**6
COMMON/ERR/S(10,10),GM(10),GN(10,10),SS(10,10),XF(10),XN( 00011*42
*10),LM(10) 00012*42
COMMON/CONT/ID,HEQ(8),RADN,NUMKP,NUMSL,NEQ,MBAND,NCYCL,NDYN,IPLNA 00013**5
*X,ISTK,K,IPRINT,ISTAR1,NEOH,MBANDH 00014*16
*NEQG,MBANDG 00015*45
COMMON/SOL/ARUK,MBLOCK,NEQB,LL,NF,LDUM,NEIG,NAD,NVV,NFO 00016**4
COMMON/ASOL/W1 00017**4
COMMON/ELPAK/NPAR(14),NELTYP 00018*11
COMMON/EXTRA/MDTA,N10,N10SV,N110 00019*11
COMMON/TAPES/NOO(6),NUMBER 00020*49
COMMON/QUAT/RF(4),ZZ(4),FAC,K,Q(6),P(4,12),SF(12,12),H(12,12), 00021*28
*S1(20,12),DG(4,4),C(12,12),E(12,12),JJ(5,12),P1(4,8) 00022*30
COMMON/BUDDY/KGS,NUMA1,NC,LOUP,NDRN,NINT,LDWA,NPT 00023*44
00024*11
00025*11
00026*11
00027*11
00028*14
DIMENSION IK(8),LH(8),SK(8,8),          GMS(8),NP(4) 00029**2
DIMENSION XMS(1),AI(NEQS,1),DBETA1(1),DBETA2(1),ID(NUMNP,1),IH(NUM 00030*11
*NP,1),DPL(1),GMX(1),GUSE(1),DBT1(1),DBT2(1),ES(1),EN(1),PU(1) 00031**6
DIMENSION AS(NEQS,1),X(NEQS,1),ROOT(NEQS),VROOT(NEQS,1),
*RUO1(NEQS)
00032**6
00033**6
REWIND 22
REWIND 61
REWIND 51
00034*11
00035*11
00036
00037*32
00038*32
00039
00040
DU 300 I=1,NEQS
XMS(I)=0.0
DU 300 J=1,MBANDS
300 AI(I,J)=0.0
00041**2
00042
00043*14
00044*14
00045*41
K=1
WRITE(6,451)
451  FORMAT(1H1/5X,' PRINT IN SUD ')
00046*14
00047
DU 100 N=1,NUMEL
READ(51)(IK(I),I=1,8),(LH(I),I=1,8),((SK(I,J),J=1,8),I=1,8)
*,((P1(I,J),J=1,8),I=1,4),(GMS(I),I=1,8),VOL,DENS,DENF,P,NP
00048*18
00049*34
00050*22
00051*22
00052*32
**** CHECK HERE FOR SHARPDOWN UNDER UNDRAINED CONDITION 00053*31

```

**** SHOULD WE USE SK OR SF **** CHANGE IN CONSV FOR 00054*31
 *** ED CONDITION BY ADJUSTING THE LAME 'S' CONSTANTS *** 00055*31

00056*31
 00057*31
 READ(51) ((SF(I,J),J=1,8),I=1,8),((R(I,J),J=1,8),I=1,8),((C(I,J),J=1,8),I=1,8),IMAT,KS,KF,ALF,((H(I,J),J=1,8),I=1,8)
 *,(XP(I),I=1,16),(XN(I),I=1,16) 00058*18
 00059*18
 00060*43

IF(KKS.EQ.2) GO TO 6000 00061*22
 00062*22

KG=IMAT 00063*32
 00064*41

IF(LOOP.FO.1) GO TO 450 00065*41
 00066*46
 00067*46
 00068*46
 00069*47

READ(22) ENEW,GNEW,DAMP,POIS 00070*47
 00071*47
 00072*47

AK=ENEW/ED(IMAT) 00073*47
 00074*46
 00075*41
 00076*41

450 CONTINUE 00077*46
 00078*46
 00079*46

IF(LOOP.EQ.1) AK=1.0

WRITE(6,452) N,AK

452 FORMAT(5X,15,210,J)

DO 400 I=1,8 00080*46
 00081*22
 00082*22

DO 400 J=1,8 00083*11
 00084*11

400 SK(I,J)=SK(I,J)*AK 00085*11
 00086*11

DO 600 I=1,8 00087*11
 00088*11

DO 600 J=1,8 00089*14
 00090*43

600 SF(I,J)=SK(I,J)+ALF*E(I,J) 00091*11
 00092*25
 00093*25
 00094*25
 00095*27
 00096*27
 00097*27
 00098*25

WRITE(61) VOL,DENS,DENF,F,SF,E,C,IMAT,KS,KF,ALF,H,NP

*,XP,XN

5000 CONTINUE

550 CONTINUE 00099
 00100
 00101
 00102**4
 00103
 00104
 00105
 00106
 00107
 00108*34
 00109*34
 00110
 00111*33

DO 300 I=1,8

LP=LK(I)

IF(LP) 360,360,290

290 XMS(LP)=XMS(LP)+GMS(I)

301 DO 350 J=1,8

KL=LK(J)

IF(KL) 350,350,250

250 IF(KL-LP) 350,390,390

390 KG=KL-LP+1

IF((KKS.EQ.4.AND.NDRN.EQ.0).OR.(KKS.EQ.2.AND.NDRN.EQ.0))

* GO TO 365

A1(LP,KG)=A1(LP,KG)+SK(I,J)

GO TO 350

```

*** SHAKE DOWN UNDER DRAINAGE CONDITION *****
365 A1(LP,KG)=A1(LP,KG)+D1(1,0)
350 CONTINUE
360 CONTINUE

KK=KG+1

100 CONTINUE
IF(KKS.EQ.4) GO TO 9000

REWIND 9
REWIND 4
NPAR(1)=1
NEB=1
NEQB=1
NELOCK=1
MM=1
NDEG=0
NVV=0
ANORM=0.0
DO 710 I=1,NEQS
IF(A1(I,1).NE.0) NDEG=NDEG+1
IF(XMS(I).NE.0) NVV=NVV+1
IF(A1(I,1).EQ.0) A1(I,1)=1.0E+20
710 CONTINUE
WRITE(4)((A1(I,J),I=1,NEQS),J=1,MBANDS)
WRITE(9)(XMS(I),I=1,NEQS)
IF(NDEG.GT.0) GO TO 730
WRITE(6,1010)
STOP
730 ANORM=(ANORM/NDEG)*1.0E-20
1010 FORMAT(51)
STRUCTURE WITH NO DEGREES OF FREEDOM CHECK DATA )

CALL SOLVIG(LOOP,1H,RDDM,NN15,NN17)
WRITE(6,105)
105 FORMAT(1H) *** PRINT OF ASSEMBLED UNDRAINED MATRIX ** )
WRITE(6,106)((A1(I,J),J=1,MBANDS),I=1,NEQS)
WRITE(6,106)(XMS(I),I=1,NEQS)
106 FORMAT(5X,0F12.5)

CALL STIFF1(A1,NEQS,MBANDS,AS)
DO 200 I=1,NEQS
DO 200 J=1,NEQS
IF(I.EQ.J) X(I,J)=XMS(I)
IF(I.NE.J) X(I,J)=0.0
200 CONTINUE
WRITE(6,101)
101 FORMAT(1H) *** PRINT OF ASSEMBLED UNDRAINED MATRIX ** )
WRITE(6,102)((AS(I,J),J=1,NEQS),I=1,NEQS)
WRITE(6,209)((X(I,J),J=1,NEQS),I=1,NEQS)
102 FORMAT(5X,0F12.5)
CALL MROU(NEQS,AS,X,KOUT,VROU)

```

```

00112*32
00113*32
00114*32
00115*32
00116*32
00117
00118
00119*25
00120*41
00121*41
00122*25
00123
00128
00129*11
00130*11
00131*11
00132
00133
00134*11
00135*11
00136
00137
00138
00139
00140
00141
00142
00143
00144
00145
00146
00147**3
00148
00149
00150
00151
00152*44
00153
00154
00155
00156
00157*25
00124*39
00125*39
00126*39
00127*40
00158
00124*39
00125*39
00126*39
00127*40

```

```

WRITE(6,209)((AS(I,J),J=1,NEWS),I=1,NEWS)
WRITE(6,209)(ROOT(N),N=1,NEWS)
209 FORMAT(1A,1E10.3)
DO 210 N=1,NEWS
ROOT1(N)=ROOT(N)
210 CONTINUE
DO 211 N=1,NEWS
NK=NEWS+1
211 ROOT(N)=ROOT1(NK-N)

DO 212 N=1,NEWS
ROOT1(N)=(DSQRT(ROOT(N)))
WRITE(6,209)(ROOT1(N),N=1,NEWS)

IF(MDIN.EQ.5) GO TO 9000

W1=ROOT1(1)

REWIND 61

IF(LOOP.NE.1) GO TO 850

DO 800 M=1,NUMMAT
DBT1(M)=DBL(M)*W1
DBT2(M)=DBL(M)/W1
800 CONTINUE

GO TO 855

850 CONTINUE

REWIND 22

DO 860 M=1,NUMEL
READ(22) ENLW,GMELW,DAMP,POIS
DBT1(M)=DAMP*W1
DBT2(M)=DAMP/W1
860 CONTINUE

855 CONTINUE

REWIND 13

DO 500 M=1,NUMEL

NEL=M
READ(61) VOL,DENS,DFNF,F,SE,E,C,IMAT,KS,KF,ALF,H,NP
*,XP,XH

```

00159

```

00160*11
00161*47
00162*47
00163*47
00164*47
00165
00166*11
00167*11
00168
00169*47
00170*47
00171*47
00172*47
00173*47
00174*47
00175*47
00176*47
00177*47
00178*11
00179*11
00180*11
00181*47
00182*47
00183*47
00184*47
00185*47
00186*47
00187*47
00188*48
00189*48
00190*48
00191*48
00192*48
00193*21
00194*25
00195*25
00196*25
00197*14
00198*43

```



```

L=0
DO 620 I=1,4
  IIM=1
  N=NP(IIM)
  DO 620 J=1,4
    L=L+1
    LM(L)=ID(N,J)
620 CONTINUE

CALL RSPDMX(VOL,DENS,DEFN,KS,KF,DBT1,DBT2,F,NUMNP,NEL,ID,IH,IMAT)
***** SHOULD WE CALL A LOAD ROUTINE *****
***** REFORMULATE FOR SELF WT *****
WRITE(13) LM,S,GM,GH,SS,XP,XN
500 CONTINUE
9000 CONTINUE

RETURN
END

```

```

00199*14
00200*14
00201*14
00202*14
00203*14
00204*14
00205*14
00206*14
00207*14
00208*14
00209*14
00210*14
00211*14
00212*26
00213*34
00214*34
00215*34
00216*42
00217*42
00218*42
00219*42
00220*42
00221*42
00222*42
00223
00224
00225*31
00226*31
00227
00228

```

```

DATA SET SLDN AT LEVEL 010 AS OF 83/02/10
SUBROUTINE SLDN (A1,D1,X0,A5,NEG,X1,X2,A4,NWA,A0,MAXA,NEQB,
*NBAND,C1,MBAND,NC,XM)
IMPLICIT REAL *B(A-H,O-Z)
COMMON/ACON/ITLAG
COMMON/ACNST/DELT1,DELT2,C1,C2,A0,A01,A02,A03,A04,A05,A06,A07,A08
COMMON/CONTR/DO,HEP(8),FADN,FASIM,NUMSL,NKK,NBAND,NCYCL,NDYN,IPLNA
*X,ITREF,NPRINT,ISJANT,NEGR,MBAND
*,NEGS,MBANDS
COMMON/BOBBY/KKS,NUMN1,NG,LOUP,NDRN,NINT,IDWA,NPT
COMMON/INTGE/ETA,BETA,GAMA,RTOL,NGAMA,NPUNCH,NEQUIB,ITEMX
DIMENSION AT(1),X0(1),X1(1),X2(1),DT(1),A5(NEG,1),A4(NWA),A0(1),
*XM(1),MAXA(1)
*****#END*****
*****DIVERT FROM THIS STEP CODED BY HALDAR *****
SOLVE FOR DISPLACEMENTS,VELOCITIES AND ACCELERATIONS
DO 4500 I=1,NEG
***** FOR INELASTIC ANALYSIS *****
GO TO (510,510,500,510),KKS
510 CONTINUE
AT(1)=A0*X0(I)+A02*X1(1)+2.*X2(1)
HT(I)=A01*X0(1)+2.*X1(1)+0.5*DELT1*X2(1)
GO TO 4500
500 AT(1)=A02*X1(1)+2.*X2(1)
HT(1)=2.*X1(1)+A03*X2(1)
4500 CONTINUE
DO 450 I=1,NEG
GO TO (610,610,600,610),KKS
610 CONTINUE

```

```

00001
00002**4
00003
00004**4
00005
00006**4
00007**8
00008**4
00009
00010
00011
00012*10
00013**9
00014**9
00015**9
00016**9
00017**4
00018**4
00019
00020
00021
00022
00023
00024
00025
00026
00027
00028**3
00029**3
00030
00031
00032
00033
00034**4
00035**4
00036**4
00037**8
00038**8
00039**8
00040**6
00041
00042
00043**5
00044**4
00045**4
00046**4
00047**4
00048
00049**3
00050**3
00051
00052
00053
00054**4
00055**4
00056**8
00057**8
00058**8
00059

```

```

AU(1)=AU(1)+AM(1)*(A0+XU(1)+A02+X1(1)+2.*X2(1))+A5(1,1)*(A01*XU(1)
*+2.*X1(1)+0.5*DEL11*X2(1))
GO TO 450
***** FOR INELASTIC ANALYSIS ONLY *****
600 AU(1)=AU(1)+AM(1)*A1(1)+A5(1,1)*B1(1)
450 CONTINUE

DO 451 N=1,NEQ
  LU=N-1
  MR=MIN0(M0AM1,NEQ-N+1)
  IF(MR.EQ.1) GO TO 3000
  DO 3200 J=2,MR
    M=LU+J
    AU(N)=AU(N)+A5(N,J)*(A01*XU(M)+2*X1(M)+0.5*DEL11*X2(M))
    AU(M)=AU(M)+A5(N,J)*(A01*XU(N)+2.*X1(N)+0.5*DEL11*X2(N))
  3200 CONTINUE
  451 CONTINUE
  3000 CONTINUE

  WRITE(6,7000) NG
  FORMAT(1H1,/,
  5X,'          PRINT IN SOLUTION 1/
  *          TIME STEP(NG) = ',15/)

  WRITE(6,7020)
  FORMAT(5X,'          MODIFIED LOAD VECTOR AU(1) 1/)

  WRITE(6,7030) (AU(I),I=1,NEQ)
  WRITE(6,7030) (A1(I),I=1,NEQ)
  WRITE(6,7030) (B1(I),I=1,NEQ)
  7030   FORMAT(5X,12F10.3)

*****
*****SPECIAL FORM FOR STATIC ANALYSIS CONSISTENT WITH KSTAR**
***** PRINT CERTAIN QUANTITIES *****

```

```

00060
00061
00062**4
00063**5
00064**4
00065**4
00066**4
00067**4
00068**4
00069**4
00070
00071**4
00072**4
00073
00074
00075
00076
00077
00078
00079
00080
00081
00082**4
00083**4
00084**4
00085**8
00086**8
00087**8
00088
00089
00090
00091**5
00092**5
00093**5
00094**4
00095**4
00096**4
00097**4
00098
00099
00100
00101**5
00102*10
00103*10
00104*10
00105*10
00106*10
00107*10
00108*10
00109*10
00110*10
00111**5
00112
00113
00114
00115
00116
00117**5

```

```

CALL RENVK(A+,AU,PA,AA,NEQ,NA,NEG,NBLOCK,#1,NBAND,NC,KKS) 00118**6
00119**5
00120**5
*****00121
00122
00123
00124**8
IF(MEQDIF.NE.0) GO TO 5540 00125**9
00126**9
00127**8
00128
00129
00130
00131
00132
00133
00134
00135
00136**8
DELT=DD 00137**8
DO 590 I=1,NEQ 00138**8
UO=XO(I) 00139
U1O=X1(I) 00140
U2O=X2(I) 00141
U=AU(I) 00142**4
GO TO (810,810,800,810),KKS 00143**9
00144**4
810 CONTINUE 00145**4
U1=AO1*U-BO(I) 00146**4
U2=AO*U-A1(I) 00147**4
X2(I)=C2*U2O+C1*U2 00148**4
00149**4
GO TO 820 00150**9
00151**4
***** FOR INELASTIC ANALYSIS ONLY MODIFY X2(I) ***** 00152**7
***** U DENOTES INCREMENTAL DISPLACEMENT BETWEEN ' T+DT ' 00153**7
***** ' I ' TIME STEP ***** 00154**7
00155**6
830 CONTINUE 00156**6
00157**9
800 X2(I)=AO4*U+AO5*U1O+AO6*U2O 00158
00159
00160
00161**8
00162**8
820 CONTINUE 00163
00164**4
X1(I)=U1O+DELT/2.*(U2O+X2(I)) 00165*10
XU(I)=UO+DELT*U1O+AO8*(X2(I)+2.*U2O) 00166*10
590 CONTINUE 00167*10
00168*10
00169*10
00170*10
5540 CONTINUE 00171*10C
00173*10
00173*10
00173*10
00173*10
00175**4
00176
RETURN 00177
END

```

```

DATA SET SOLFIG AT LEVEL 013 AS OF 83/02/16
SUBROUTINE SOLFIG(LOOP, IH, NDIM, NN15, NN17)
IMPLICIT REAL*8(A-H, O-Z)
CALLS: MODES, PRINTD
CALLED BY: MAIN
SOLUTION OF THE EIGENVALUE PROBLEM
COMMON A(1)
COMMON/EI.PAR/NPAR(14), NELIYI
COMMON/CONT/DT, RED(8), RADN, NOMNF, NOMEI, NEG, MBAND, NCYCL, NDYN, IPLNA
*X, ISTRPR, NPRINT, ISTART, NEQH, MBANDH, NEQS, MBANDS
COMMON/QUAD/CCCC(1500), N1, N2, N3, N4, N5, MTUT, NNN
COMMON /SOL/ANORM, NBLUCK, NEQB, LI, NF, IDUM, NEIG, NAD, NVV, NFO
COMMON /EM/ AT(1000), IFILL1(3136)
COMMON /EXTRA/ MDEX, N18, IFILL2(14)
COMMON/ASOL/W1
DIMENSION IH(NOMNF, 1)
REAL*4 I(3)
DIMENSION C(100)
*****CHANGED *****
NI=8
READ CONTROL CARD
REWIND 14
WRITE (6,1003)
IF (LOOP.NE.1) GO TO 101
READ (5,100) IPRR, IFSS, NITEM, RTOL, CUFQ, NFO
WRITE(14) IPRR, IFSS, NITEM, RTOL, CUFQ, NFO
GO TO 102
101 READ(14) IPRR, IFSS, NITEM, RTOL, CUFQ, NFO
102 IF (IPRR.GT.0) IPRR=1
IF (IFSS.GT.0) IFSS=1
IF (NITEM.EQ.0) NITEM=16
IF (RTOL.EQ.0.) RTOL=1.E-05
IF (CUFQ.EQ.0.) CUFQ=1.108
IF (NEIG.GT.0) GO TO 10
WRITE (6,1001)
GO TO 15
10 WRITE (6,1002)
15 WRITE (6,1000) IPRR, IFSS, NITEM, RTOL, CUFQ, NFO
20 IF (MDEX.EQ.1) RETURN
TPI=8.000+DATAN(1.000)
CUFQ=CUFQ*TPI
CUFQ=CUFQ*CUFQ
CALL SOLUTION ROUTINE
CALL MODES(NEQS, MBANDS, NBLUCK, NEQB, NF, MTUT, IPRR, IFSS, RTOL, NITEM,
*CUFQ, NN15, NN17)
WRITE CONTROL INFORMATION ON TAPE -- FOR RESTART OPTION

```

```

00001**9
00002
00003
00004
00005
00006
00007
00008
00009**6
00010
00011
00012**4
00013*12
0014
00015
00016
00017
00018
00019*12
00020
00021*13
00022
00023*11
00024
00025
00026
00027
00028
00029
00030
00031
00032
00033
00034
00035
00036
00037
00038
00039
00040
00041
00042
00043
00044
00045
00046
00047
00048
00049
00050
00051
00052
00053
00054**4
00055**9
00056
00057
00058
00059

```

	NC=2	00060
	REWIND NC	00061
	WRITE (NC) NREQ,NBLOCK,NEQB,NBAND,N1,NF,(A1(I),I=1,NF)	00062
	PRINT OF EIGENVALUES (OMEGA) AND EIGENVECTORS	00063
		00064
	REWIND N1	00065
	READ (N1) (C(I),I=1,NF)	00066
	K=NF+1	00067
	DO 30 I=1,NF	00068
	K=K-1	00069
	KK=(K-1)*J+1	00070
	C(KK)=C(K)	00071
	C(KK+1)=C(K)/IP1	00072
	C(KK+2)=IP1/C(K)	00073
30	IF (HE10.01.0) GO TO 25	00074
	WRITE (0,1009)	00075
	DO 41 I=1,NF	00076
	K1=J*1-2	00077
	K2=J*1	00078
41	WRITE (0,1020) 1,(C(J),J=K1,K2)	00079
	GO TO 35	00080
25	WRITE (0,1010)	00081
	DO 40 I=1,NF	00082
	K1=J*1-2	00083
	K2=J*1	00084
40	WRITE (0,1020) 1,(C(J),J=K1,K2),A1(NF+1)	00085
		00086
	*****CHANGED *****	00087
35	CONTINUE	00088
	W1=C(2)	00089**5
	NN16=NN15+2*NF	00090**8
	NN17=NN16+NEQB*NF	00091*11
		00092*12
		00093*11
		00094*11
	WRITE (0,1030)	00095
	REWIND 14	00096
	READ(14) IFPK,IFSS,NITEM,KTOL,COPU,NFO	00097
	WRITE(0,98) ((IH(I,J),J=1,2),I=1,NUMNP)	00098**7
98	FORMAT(5X,215)	00099**9
	CALL PRINTD(IH,A(NN15),A(NN16),NEQB,NUMNP,NF,NBLOCK,NEQB,NT,2)	00100**9
		00101
	COMPUTE TIME LOG	00102
		00103
		00104
		00105
		00106
		00107
100	FORMAT (315,2F10.0,15)	00108
1000	FORMAT (10H // 20H CONTROL INFORMATION, //	00109
1	5X,31HFLAG FOR ADDITIONAL PRINTING =, 15 /	00110
2	7X,14HEQ.0, SUPPRESS, /	00111
3	7X,11HEQ.1, PRINT, //	00112
4	5X,31HFORM SEQUENCE CHECK FLAG (*) =, 15 /	00113
5	7X,19HEQ.0, PERFORM CHECK, /	00114
6	7X,10HEQ.1, PASS, //	00115
7	5X,31HMAXIMUM ITERATION CYCLES (*) =, 15 //	00116
8	5X,31HCONVERGENCE TOLERANCE (*) =, E14.4 //	00117
9	5X,31HCUT-OFF FREQUENCY (CPS) =, E14.4 //	00118
*	5X,31HNUMBER OF STARTING ITERATION , /	00119
*	5X,31HVECTORS TO BE READ FROM , /	00119
*	5X,31HAPR10 (*) =, 15 ///	00119


```

DATA SET SOLST      AT LEVEL 011 AS OF 83/02/02
SUBROUTINE  SOLST(A4,AU,NEQ,MBAND,X0,NC,NWA,NEGB,NBLOCK,M1,MAXA) 00001
IMPLICIT REAL*8 (A-H,O-Z) 00002**3
COMMON/GORA/IFLAG,IR 00003
COMMON/INTEGE/TETA,BETA,GAMA,RTOL,NGAMA,NPUNCH,NEQUIB 00004**3
COMMON/BOBBY/KKS,NORMAT,NG,LOOP,NDRN,NINT,IDWA,NPT 00005**9
DIMENSION X0(1),A4(NWA),MAXA(1),AU(1) 00006**9
*****SPECIAL FOR STATIC ANALYSIS*****00007**9
CALL REUVK(A4,AU,MAXA,NEGB,NWA,NEQ,NBLOCK,M1,MBAND,NC,KKS) 00008*11
GO TO (1,1,300,1),KKS 00009
1 CONTINUE 00010
DO 100 I=1,NEQ 00011**3
X0(I)=AU(I) 00012**3
GO TO 900 00013**7
00014**7
00015**6
00016**8
00017**8
00018**8
00019**8
00020**8
00021**8
00022**8
00023**8
00024**8
00025**8
00026**8
00027**8
00028**6
00029**7
00030**6
00031**6
00032**9
00033**9
00034**9
00035**4
00036**4
00037**5
00038**4
00039**4
00040
00041**3
RETURN 00042
END 00043

```


DATA SET SUBLIN	AT LEVEL 005 AS OF 83/02/11	00001
SUBROUTINE	SUBIN(A1,B1,X0,A5,NEQ,X1,X2,A4,NWA,A0,MAXA,NEQB,	00002**3
*NBLUCK,M1,MBAND,NC)		00003
IMPLICIT REAL *8 (A-H,O-Z)		00004**3
		00005
COMMON/CONTL/DO,NEQ(N),KALN,NAS14,NUMSD,MRK,MBAND,NCYCL,NDYN,IPLNA		00006**3
*X,ISTRPR,NPRINI,ISTART,NEQK,MBANDN		00007
COMMON/INTEG/TETA,BETA,GAMA,RTOL,NGAMA,NPUNCH,NEQUIB		00008
COMMON/BOBBY/KKS,NUMMAT,NG,LOOP,NDRN,NINT,IDWA,NPT		00009**5
		00010**5
DIMENSION A1(1),BT(1),X0(1),A5(NEQ,1),X1(1),X2(1),A4(NWA),A0(1),		00011**5
*MAXA(1)		00012
		00013
		00014**3
		00015**3
DT=TETA*DD		00016
DO 4500 I=1,NEQ		00017
4501 AT(1)=0.0		00018
BT(1)=(1./DT)*X0(1)		00019
4500 CONTINUE		00020
DO 454 I=1,NEQ		00021
459 AU(1)=AU(1)+A5(I,1)*BT(1)		00022
DO 451 N=1,NEQ		00023
LQ=N-1		00024
MR=MIN0(MBAND,NEQ-N+1)		00025
IF(MR.NE.1) GO TO 3600		00026
DO 3200 J=2,MR		00027
M=LQ+J		00028
3201 AU(N)=AU(N)+A5(N,J)*BT(M)		00029
AU(M)=AU(M)+A5(N,J)*BT(N)		00030
3200 CONTINUE		00031
451 CONTINUE		00032
3600 CONTINUE		00033
9000 CONTINUE		00034
8000 CONTINUE		00035
CALL REDVR(A4,A0,MAXA,NEQB,NWA,NEQ,NBLUCK,M1,MBAND,NC,KKS)		00036**4
		00037**4
5540 DO 5900 I=1,NEQ		00038
U0=X0(1)		00039
U10=X1(1)		00040
U=A0(1)		00041
X1(J)=(1./DT)*(U-U0)		00042
X0(1)=U		00043
5900 CONTINUE		00044
		00045**3
		00046**3
RETURN		00047
END		00048

```

DATA SET SOLVI      AT LEVEL 051 AS OF 83/03/05
SUBROUTINE SOLVI (IH,A1,A3,XM,B,BU,VEL,PDYL,IS,GACL,LD,R,A,AU, 00001
                                00002**9
*GB,PMAX,AMAX,NEG,NUMNP,NACL,NUMEL,NERH,GCB,A4,A5,A6,MAXA,X0,X1,X2,00003*30
                                00004**9
*A1,BT,AC, SU,NWA,NEQB,NBLUCK,M1,X,Y,Q,ISC,JSC,SURTRX,SURTRY,AM,EK,00005
                                00006**9
*AN,NSLC1,ANI,ERJ,DC,SURPFX,SURPFY,RE,SW,AS) 00007*20
                                00008**9
IMPLICIT REAL*8 (A-H,O-Z) 00009
                                00010**3
                                00011**3
                                00012**3
COMMON/CUNIT/DD,HEB(8),KADN,NASIM,NUMSL,NKK,MBAND,NCYCL,NDYN,IPLNA00013
*X,ISTRK,NPRINT,ISJARI,NEQB,MBANDH,NEQS,MBANDS 00014**3
COMMON/NASC/NTP,NSC,NSIC,NLC,NNYP,NWSC,ISIC 00015*13
COMMON/SOL/ANRM,NGLOCK,NEQB,NDC
***** COMMON BLOCK INSERTED FOR CHECKING ****
COMMON/BEA/NS,NELTYP,ND,NUMGT,MODEL,NCUN,LDW,NK015,NK016,NK017
*****
COMMON/INT/GE/ETA,BETA,GAMA,KYDL,NGAMA,NPUNCH,NEQUB,ITEMX 00016*23
COMMON/BOBBY/KKS,NUMMAT,NC,LDUP,NDRN,NINT,LDWA,NPT 00017*23
COMMON/ACNST/DEL1,DEL2,C1,C2,A0,A01,A02,A03,A04,A05,A06,A07,A08 00018*41
COMMON/PRIM/ISIKS,MPRINT,JPRINT,KPRINT,IP 00019
COMMON/SURK/ITL6,ITR 00020*45
                                00021*47
COMMON/DINEL/N43,IP1,IP11,KK 00022*47
                                00023*27
COMMON/EL/IND,ICOUNT 00024**3
                                00025*33
                                00026*33
                                00027**3
                                00028**3
DIMENSION A1(NEQB,1),GCB(1),A3(NEQ,1),XM(1),B(1),BU(1),VEL(1), 00029*30
*PDYL(2,4,1),IS(1),GACL(1),LD(2),HEQB(8),LD(NUMNP,1),R(1),A(1), 00030
*AU(1),GB(1),PMAX(NUMEL,1),AMAX(NUMNP,12,2),DS(4,3),IH(NUMNP,1) 00031*45
                                00032**3
DIMENSION AS(NEQS,1) 00033**3
DIMENSION A4(NWA),A5(NEQ,1),A6(NWA),MAXA(1),X0(1),X1(1),X2(1),AT( 00034**3
*1),BT(1),AC(1),SU(1),X(1),Y(1),Q(1),ISC(1),JSC(1),SURTRX(2,ND 00035
*SURTRY(2,ND,2),AN(NSC,1),AM(1),EK(1),ANI(NYP,1),ERJ(NYP),NSLC1(1 00036
*) ,SURPFX(2,ND,2),SURPFY(2,ND,2) 00037**6
                                00038**2
                                00039**3
DIMENSION RE(1) 00040*10
DIMENSION SW(1) 00041*20
                                00042**3
COMMON/TAPES/NS1F,NKED,BL,NR,IFIL(2),NUMBER,ICOUNT 00043*49
                                00044
                                00045
***** TEMPORARY STORAGE FOR LOAD VECTOR ADDITION ***** 00046
                                00047*32
                                00048*32
                                00049*32
                                00050*32
                                00051*32
                                00052
IND=-1
IF(KKS.EQ.3.AND.IND.EQ.-1) GO TO 500
CONTINUE

```

```

NC=0
DT=0.0
IND=NC
TR=NCYCL*DD
TM=0.0
00053
00054
00055
00056*32
00057*32
00058
00059**9
00060**9
00061**9
***** BRANCH FOR WORKING ARRAY INITIALIZATION IN ELT2D6 **00062**9
00063**9
00064**9
***** INITIALIZE GLOBAL LOAD VECTOR *****
00065
700 DO 1050 I=1,NEQ 00066
1050 A0(I)=0.0 00067
00068*11
*****
00069*16
00070*19
WRITE(6,100)
100 FORMAT(5X, ' PRINT OF BETA IN SOLVI BEFORE ELEMENT ')
WRITE(6,105) NS,NELTYP,ND,NUMGT,MODEL,NCON,LDW,NK015,NK016,NK017
105 FORMAT(5X,10I5)
*****
***** ELASTOPLASTIC ANALYSIS START 00071*19
00072*11
00073*11
00074*11
IF(NKS.NE.3) GO TO 600 00075*11
IF(NC.EQ.0) GO TO 600 00076*12
00077*11
CALL ELEM1(NUMMAT) 00078*11
00079*11
CALL KSTAR(LH,A1,A3,XM,NEQ,MBAND,NEQH,MBANDH,A4,A5,A6,MAXA, 00080*11
00081*21
*NWA,NEQD,NBLOCN,M1,LD,KE,DW) 00082*21
*****
WRITE(6,105) NS,NELTYP,ND,NUMGT,MODEL,NCON,LDW,NK015,NK016,NK017
600 CONTINUE
00083*11
00084*11
00085*11
00086*11
00087
00088
00089
***** ASSEMBLE LOAD VECTOR***
00090**7
00091**7
00092**7
00093**7
00094*48
00095*48
00096
CALL LOAD1(NUMNP,NEQ,LC,ISLC,DT,NC,ISC,JSC,SURTRX,SURTRY,X,Y, 00097**6
00098**6
*,LD,SD,TR,TM,PDTL,SURFFX,SURFFY)
00099
00100
00101**8
DO 1000 I=1,NEQ
1000 R(I)=0.0 00102**8

```

```

CALL LOADZ(FUNDP,FDYL,A,FUNLP,LD,DI,IO,NC,IR,R)
1 CONTINUE
  DO 5 I=1,NEQ
  AU(I)=SU(I)+R(I)
2 CONTINUE
6 CONTINUE
  GO TO (51,51,51,52),KKS
52 CONTINUE
***** ONLY FOR LIMIT ANALYSIS PROBLEM *****
  IF(IPT1.EQ.2.AND.LC.EQ.NLC) GO TO 55
  IF(IPT1.EQ.1.AND.LC.EQ.NLC) GO TO 55
51 CONTINUE
***** SAVE LOAD VECTOR FOR EACH CYCLE BEFORE ITER *****
  DO 49 I=1,NEQ
49 GB(I)=AU(I)
  IF(KKS.NE.3) GO TO 55
***** CALCULATE NORM OF THE ACTUAL INCREMENTAL LOAD APPLIED ***
  RNORM=0.
  DO 50 I=1,NEQ
  RNORM=RNORM+GB(I)*GB(I)
50 CONTINUE
800 DO 800 I=1,NEQ
  AU(I)=SU(I)+R(I)-RE(I)
55 CONTINUE
  IF(NC.EQ.0) GO TO 10

```

```

00103**6
00104
00105*37
00106*37
00107*37
00108*37
00109*37
00110*37
00111*37
00112*37
00113*43
00114*27
00115*27
00116*37
00117*37
00118*37
00119*38
00120*38
00121*38
00122*38
00123*38
00124*38
00125*38
00126*27
00127*27
00128*39
00128*39
00129*16
00130*20
00131*38
00132*38
00133*20
00134*20
00135*20
00136*20
00137*20
00138*28
00139*30
00140*22
00141*39
00142*39
00143*39
00144*22
00145*22
00146*25
00147*25
00148*25
00149*25
00150*30
00151*25
00152*25
00153*25
00154*43
00155*43
00156*22
00157*20
00158*39
00159*20
00160*50
00161*50

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```

                                00162*50
LCOUNT=LCOUNT+1
WRITE(6,105) LCOUNT

                                00163*51
                                00165*51
                                00167*50
                                00168*50
                                00169*50
                                00170*50
IF(LCOUNT.NE.151RFR) GO TO 20

10 CONTINUE

                                00171*50
                                00172*16
                                00173*16
45  FURMAT(1H)/' *** PRINT OF ASSEMBLED LOAD VECTOR IN SOLV1 ' 00174*16
WRITE(6,40) (AU(I),I=1,NEQ) 00175*16
46  FURMAT(5X,9E12.5) 00176*16
20  CONTINUE 00177*6
                                00178*50
                                00179*35
                                00180*43
                                00181*43
43  IF(NC.NE.0) GO TO 42 00182*43
DO 43 I=1,NEQ 00183*43
42  AU(I)=AU(I)-SW(I) 00184*43
CONTINUE 00185*6
                                **00186*11
                                00187*11
                                00188*44
                                00189*6
                                00190
                                00191
                                00192
                                00193
                                00194*20
                                00195*44
                                00196*20
                                00197*21
                                00198*21
                                00199*25
                                00200*25
                                00201*25
                                00202*25
                                00203*25
                                00204*21
                                00205*39
                                00206*39
                                00207*23
                                00208*23
                                00209*32
***** CHECK OVERALL LOGIC FOR ALL MATERIAL MODEL
IF(NC.EQ.0) GO TO 41
15TRES=15TRES+1
MPRINT=MPRINT+1
JPRINT=JPRINT+1
KPRINT=KPRINT+1
41  CONTINUE
***** PROCEED FOR ITERATION *****
***** SET ITERATION NUMBER *****
*****
***** IFLAG. EQ. 0 MEANS NOT CONVERGED VALUES *****
***** IFLAG. EQ. 1 MEANS CONVERGED VALUES *****
IFLAG=0
NITR=176.FX
ITR=0
WRITE(6,105) NC,NEQ,ITR,NB,NUMGT,MODEB,NEON,10W,NK015,NK016,NK017

IF(NC.NE.0) GO TO 440
CALL SYMSUB(1,A3,AU,NEQ,MBAND)
CALL SYMSUB(2,A3,AU,NEQ,MBAND)

IF(NRS.EQ.3) GO TO 451
                                00210*32
                                00211*32
                                00212*32
                                00213*32
                                00214*39
                                00215*39
                                00216*39

```

		00217*39
450	DU 450 I=1,NEQ XU(I)=AU(I)	00218*32 00219*32
		00220*39
451	GO TO 420 CONTINUE	00221*43 00222*39 00223*43
	IF(NEQEQ.0) GO TO 453	00224*43
452	DU 452 I=1,NEQ XU(I)=XU(I)+AU(I)	00225*43 00226*43 00227*43
453	CONTINUE	00228*43
440	GO TO 420 CONTINUE	00229*39 00230*32 00231*32
	CALL SOLST(A4,AU,NEQ,MBAND,XU,NC,NWA,NEQB,NBLUCK,M1,MAXA)	00232*32 00233*32 00234*23 00235*23
420	CONTINUE	00236*23 00237*23 00238*32 00239*39
	WRITE (6,105) NS,NEQEQ,ND,NUMGP,MODEL,NEQN,LDW,NK015,NK016,NK017	
		00240*39
	IF(NEQEQ.0) GO TO 430	00241*23 00242*23 00243*21 00244*42 00245*21
	***** COUNTER FOR DYNAMIC EQUILIBRIUM *****	00246*21 00247*21 00248*21
		00249*20 00250*20 00251*20 00252*24
	CALL DEFCTR(XU,NDYN,AU,RE,R,ITR,NITR,ITEMA,RTOL,NC,A4,NEQ,	00253*24 00254*24
	*MBAND,NWA,NEQB,NBLUCK,M1,MAXA,X2,X1,VEL,AC,B,LD,AMAX,PMAX,AM,EK,AN	00255*34 00256*24
	* AN1,EKJ,IM,NUMGP,NUMEI,RNORM,XM,AS,GB,AT,B1,	00257*38 00258*38
	* A3,LC)	00259*38 00260*38 00261*23 00262*23 00263*23 00264*23
		00265*19 00266*22 00267*22
430	CONTINUE	00268*22 00269*22 00270*22 00271*22 00272*22
	IFLAG=1	

		00273*22
		00274*22
		00275*23
500	CONTINUE	00276*23
	CALL	00277*19
	(U1(X1,X0,X2,VEL,AC,AU,B,1D,DS,AMAX,PMAX,NUMNP,NUMEL,	00278
	*NC,AM,ER,AM,AMI,EKJ,IM,LC,NEQ,KE,ITR)	00279**6
	IF(IND.EQ.-1) GO TO 400	00280*22
	IF(NC.EQ.0) DT=TETA*DD	00281*20
	IF(NC.EQ.0) IM=IM+DT	00282*32
		00283*44
		00284*32
	IF(NC.EQ.0) GO TO 15	
	IF(LCOUNT.NE.1STPR) GO TO 30	00285*32
	LCOUNT=0	00286*50
		00287*50
		00288*50
		00289*42
15	CONTINUE	
	CALL EQUCHK(XM,AS,B,NEQ,MHAND,X0,X1,X2,R,NC,A3,NDYN,GB,AU,AS)	00290*42
		00291*44
		00292*42
		00293*42
		00294*32
30	CONTINUE	00295*50
	NC=NC+1	00296*50
		00297*37
		00298*37
	***** EQUILIBRIUM CHECK FOR EVERY LOAD STEP ****	00299**6
	IF(NC-NCYCL)700,700,900	00300*39
		00301*39
		00302*43
		00303*22
		00304*22
		00305*22
900	CONTINUE	00306
	RETURN	00307
		00308*22
		00309*22
2010	FORMAT(//// 5TH EQUILIBRIUM ITERATION IN TIME STEP = ,15//	00310*22
*	37H NUMBER OF ITERATIONS = ,15 /)	00311*22
2020	FORMAT(////46H ITERATION LIMIT REACHED S T O P OF SOLUTION)	00312*22
		00313*26
2050	FORMAT(////70H OUT-OF-BALANCE LOADS LARGER THAN INCREMENTAL LOADS	00314*26
	*AFTER ITERATION = ,15/)	00315*26
		00316*26
		00317*22
	END	00318

```

DATA SET SOLV2      AT LEVEL 032 AS OF 83/03/05
SUBROUTINE SOLV2 (IH,A1,A3,XM,B,BU,VEL,PDYL,IS,GACL,LD,R,A,AU, 00001
                                00002*11
*GB,PMAX,AMAX,NEG,NUMNP,NACL,NUMEL,NEQH,GCB,A4,A5,A6,MAXA,X0,X1,X2,00003*20
                                00004*11
*AI,BT,AC,SI,NWA,NEQB,NBLDCK,M1,X,Y,Q,ISC,JSC,SURTRX,SURTRY,AM,EK,00005
                                00006*11
*AN,NSLCI,ANI,ERJ,LC,SURFFX,SURFFY,KE,SW,AS) 00007*18
                                00008*11
IMPLICIT REAL*8      (A-H,O-Z) 00009
                                00010**3
                                00011**3
                                00012**3
COMMON/CONT/DP,HED(8),RADN,NASIM,NUMSL,NKK,MBAND,NCYCL,NDYN,IPLNA00013
*X,ISTRPK,NPRINT,ISTART,NEQK,MBANDH,NEWS,MBANDS 00014**3
COMMON/IFLEGE/IFLA,BETA,GAMA,KTOD,NGAMA,NPUNCH,NEQUIB,ITEMX 00015*21
COMMON/BOBBY/KKS,NUMMAT,NC,LOOP,NDRN,NINT,IDWA,NPT 00016*27
COMMON/GURA/IFLAG,IIR 00017*27
                                00018*27
COMMON/ACNST/DELT1,DELT2,C1,C2,A0,A01,A02,A03,A04,A05,A06,A07,A08 00019
COMMON/SUB/ANORM,NGLOCK,NEQI,NDC
                                00020*14
COMMON/ASIM/NYP,NSE,NSIC,NLC,NNYP,NNSC,ISLC 00021*14
                                00022**3
COMMON/EL/IND,ICOUNT 00023*22
                                00024*22
                                00025**3
DIMENSION AI(NEWS,1),GCB(1),A3(NEQ,1),XM(1),B(1),BU(1),VEL(1), 00026*20
*PDYL(2,4,1),TS(1),GACL(1),TD(2),HEDG(8),ID(NUMNP,1),R(1),A(1), 00027
*AU(1),GB(1),PMAX(NUMEL,1),AMAX(NUMNP,12,2),DS(4,3),IH(NUMNP,1) 00028*29
                                00029**3

DIMENSION AS(NEWS,1)
DIMENSION A4(NWA),A5(NEQ,1),A6(NWA),MAXA(1),X0(1),X1(1),X2(1),AT(00031
*1),BT(1),AC(1),SI(1),X(1),Y(1),Q(1),ISC(1),JSC(1),SURTRX(2,NDC,2),00032
*SURTRY(2,NDC,2),AN(NNSC,1),AM(1),EK(1),ANI(NYP,1),ERJ(NYP),NSLCI(1 00033**5
*),SURFFX(2,NDC,2),SURFFY(2,NDC,2) 00034**2
                                00035**3
DIMENSION RE(1) 00036*12
                                00037**3
DIMENSION SW(1) 00038*18
COMMON/TAPES/NSTIF,NRED,NL,NR,IFIL(2),NUMBER,LCOUNT 00039*30
COMMON/PRINT/ISTRES,NPRINT,OPRINT,OPRINT,IF 00040*29
                                00041
***** TEMPORARY STORAGE ***** 00042
IND=-1 00043*21
IF(KKS.EQ.3.AND.IND.EQ.-1) GO TO 600 00044*21
400 CONTINUE 00045*21
                                00046
NC=0 00047**2
TR=NC*CL*DD 00048
DT=0.0 00049*21
IND=NC 00050*21
                                00051*15
                                00052*15
                                00053*30
                                00054*15
                                00055
                                00056
***** INITIALIZE GLOBAL LOAD VECTOR *****

```

00030**3

700	DO 1050 I=1,NEQ	00057
1050	AO(I)=0.0	00058
		00059
		00060*21
	IF(KKS.NE.3) GO TO 860	00061*21
	IF(NC.EQ.0) GO TO 860	00062*21
861	CONTINUE	00063*21
		00064*23
	CALL ELEM1(NUMMAT)	00065*23
	CALL KSTAR(IH,A1,A3,XM,NEG,MBAND,NEQH,MBANDH,A4,A5,A6,MAXA,NWA	00066*21
		00067*21
	* ,NEQB,NBLOCK,M1,ID,RE,SW)	00068*23
		00069*23
860	CONTINUE	00070*23
		00071*21
		00072*21
		00073*21
		00074
	DO 9000 I=1,NEQ	00075**7
9000	SU(I)=0.0	00076**7
	***** ASSEMBLE LOAD VECTOR***	00077
		00078
	CALL LOAD1(NUMNP,NEG,LC,ISLC,DT,NC,ISC,JSC,SURTRX,SURTRY,X,Y,	00079
	* ,ID,SU,TK,TH,PDYL,SURPFX,SURPFY)	00080**5
		00081**5
		00082
		00083
	DO 1000 I=1,NEQ	00084**8
1000	R(I)=0.0	00085**8
	CALL LOAD2(NUMNP,PDYL,X,NUMLP,ID,DT,TD,NC,TK,R)	00086**5
		00087
		00088
		00089
		00090*27
1	CONTINUE	00091*27
		00092
	DO 800 I=1,NEQ	00093
800	AO(I)=SU(I)+R(I).	00094**6
		00095**5
		00096*26
		00097*26
		00098*26
		00099*26
		00100*26
		00101*26
		00102**5
		00103*21
		00104*21
		00105*26
		00106*27
	DO 49 I=1,NEQ	00107*21
49	GB(I)=AO(I)	00108*21
	IF(KKS.NE.3) GO TO 3	00109*29
		00110*29
	RNORM=0.0	00111*21
	DO 50 I=1,NEQ	00112*21
50	RNORM=RNORM+GB(I)*GB(I)	00113*21
		00114*21
		00115*21
		00116*27

```

***** RESIDUAL LOAD VECTOR CORRECTION ON CONVERGED ST00117*27
      DO 850 I=1,NEQ                                00118*27
850      AU(I)=AU(I)-RE(I)                            00119*27
                                                    00120*27
                                                    00121*27
3          CONTINUE                                00122*27
      IF(NC.EQ.0) GO TO 10                          00123*27
                                                    00124*31
                                                    00125*31
                                                    00127*32
      IF(LCOUNT.NE.1STRPK) GO TO 20              00129*31
                                                    00130*31
                                                    00131*31
10 CONTINUE                                        00132*31
                                                    00133*31
                                                    00134**5
42 WRITE(6,42)                                     00135*15
      FORMAT('1H1/' ** PRINT OF ASSEMBLED LOAD VECTOR IN SOLV2 ') 00136*15
      WRITE(6,46) (AU(I),I=1,NEQ)                 00137*15
                                                    00138*15
                                                    00139*15
20 CONTINUE                                        00140*15
                                                    00141*31
      IF(NC.NE.0) GO TO 43                          00142*27
      DO 44 I=1,NEQ                                00143*27
44      AU(I)=AU(I)-SW(I)                          00144*27
43      CONTINUE                                    00145*27
                                                    00146*27
                                                    00147*15
46 FORMAT(5X,10E12.5)                             00148*16
      ***** ELASTOPLASTIC PART NOT COMPLETE 00149*16
      IF(NC.NE.0) GO TO 45                          00150*13
                                                    00151*27
      ISTRES=ISTRES+1                              00152*27
      MPRINT=MPRINT+1                              00153*13
      JPRINT=JPRINT+1                              00154
      KPRINT=KPRINT+1                              00155
45 CONTINUE                                        00156
                                                    00157
      IFLAG=0                                       00158*27
      NITR=ITEMX                                    00159*27
      ITR=0                                         00160*27
                                                    00161*27
                                                    00162*27
                                                    00163*27
                                                    00164*27
                                                    00165**5
                                                    00166**5
                                                    00167**5
      IF(NC.NE.0) GO TO 500                          00168**2
      CALL SYMSOL(1,A3,AU,NEQ,MBAND)                00169**2
      CALL SYMSOL(2,A3,AU,NEQ,MBAND)                00170**2
                                                    00171*17
      IF(KKS.EQ.3) GO TO 451                        00172*27
                                                    00173*27
                                                    00174*17
      ***** KKS.EQ.3 AU(1) = INCREMENTAL ***** 00175*26
      ***** KKS.NE.3) AU(1) = TOTAL *****       00176*26

```

```

DO 47 I=1,NEQ
47 X0(I)=A0(I)
   GO TO 650

451 CONTINUE
   IF(NEQU1B.NE.0) GO TO 601
   DO 452 I=1,NEQ
452 X0(I)=X0(I)+A0(I)
   GO TO 650

601 CONTINUE

C***** SOLVE THE SIMPLYTENEEDUS EQUATION FOR DISPLACEMENT VEC00199
500 CONTINUE
   CALL SOLTN(AT,BT,X0,A5,NEQ,X1,X2,A4,NWA,A0,MAXA,NEQB,
   *NBLUCK,M1,MBAND,NC)

650 CONTINUE

   IF(NEQU1B.EQ.0) GO TO 600
   CALL DEQ1TR(X0,NDYN,A0,RE,R,ITR,NITR,ITEMX,RTOL,NC,A4,NEQ,
   *MBAND,NWA,NEQB,NBLUCK,M1,MAXA,X2,X1,VEL,AC,B,ID,AMAX,PMAX,AM,EK,AN
   *ANI,EKJ,IN,NUMNP,NUMEL,RNDRM,XM,A5,GB,AT,BT,A3,LC)

600 CONTINUE
   IFLAG=1

   CALL OUT(X1,X0,X2,VEL,AC,A0,B,ID,DS,AMAX,PMAX,NUMNP,NUMEL,
   *NC,AM,EK,AN,ANI,EKJ,IM,LC,NEQ,RE,ITR)

***** CHECK FOR TRANSIENT PROBLEM TIME STEP CONSIDER 0
   IF(IND.EQ.-1) GO TO 400

```

```

00177*26
00178*17
00179*17
00180*17
00181*27
00182*27
00183*27
00184*27
00185*27
00186*27
00187*27
00188*27
00189*27
00190*27
00191*27
00192*27
00193*24
00194*24
00195*21
00196*21
00197*21
00198*21
00199
00200**2
00201
00202**5
00203
00204
00205
00206
00207*27
00208*21
00209*21
00210*21
00211*21
00212*23
00213*23
00214*23
00215*23
00216*22
00217*27
00218*27
00219*21
00220*21
00221*21
00222*21
00223**2
00224*22
00225*22
00226*22
00227
00228**5
00229*27
00230*27
00231*12
00232*12
00233*12
00234*21
00235*21
00236*12

```

	IF(NC.EQ.0) DT=ETA*DD	00237*28
	IF(NC.EQ.0) TM=1M+DI	00238*21
		00239*21
	IF(NC.EQ.0) GO TO 35	
	IF(LCOUNT.NE.(SIKPR) GO TO 30	00240*31
		00241*31
	LCOUNT=0	00242*31
35	CONTINUE	00243*21
	CALL ECHK(XM,AS,B,NEG,MBAND,XU,X1,X2,R,NC,A3,NDYN,GB,A0,AS)	00244*27
		00245*28
		00246*27
		00247*27
		00248**5
30	CONTINUE	00249*31
	NC=NC+1	00250*26
		00251*26
		00252*26
		00253**5
		00254*19
		00255*19
	IF(NC-NCYCL) 700,700,900	00256*19
		00257*19
900	CONTINUE	00258
	RETURN	00259
	END	00260

```

DATA SET SOLV3      AT LEVEL 028 AS OF 83/03/05
SUBROUTINE SOLV3 (IH,A1,A3,XM,B,BO,VEL,PDYL,TS,GACL,LD,R,A,AU,    00001
                                00002*10
*GB,PMAX,AMAX,NEQ,NUMNP,NACL,NUMEL,NEGH,GCB,A4,A5,A6,MAXA,X0,X1,X2,00003*16
                                00004*10
*AT,BT,AC,  SO,NWA,NEQB,NBLOCK,MI,X,Y,Q,ISC,JSC,SURTRX,SURTRY,AM,EK,00005
                                00006*10
*AN,NSLCI,ANI,EKJ,LC,SURPFX,SURPFY,RE,SW,AS)                    00007*15
                                00008*10
IMPLICIT REAL*8      (A-H,O-Z)                                00009
                                00010**3
                                00011**3
COMMON/CONTL/DD,HED(8),RADN,NASIM,NUMSL,NKK,MBAND,NCYCL,NDYN,IPLNA00012
*X,ISTRPK,NPRIM1,ISTART,NEGR,MBANDH,NEQS,MBANDS                00013**3
COMMON/INTEGE/TETA,BETA,GAMA,RTOL,NGAMA,NPUNCH,NEQUIB,IEMX    00014*17
COMMON/BOBBY/KKS,NUMMAT,NC,LOOP,NDRN,NINT,LDWA,NPT            00015*23
COMMON/GORA/IFLAG,IIR                                         00016*23
COMMON/ACNST/DELTA1,DELTA2,C1,C2,AU,A01,A02,A03,A04,A05,A06,A07,A08 00017
                                00018*14
COMMON/ASIM/NYP,NSC,NSLC,NLC,NNYP,NNSC,ISLC                  00019*14
COMMON/SOL/ANORM,NGLUCK,NEQD,NDC                              00020**3
                                00021**3
DIMENSION A1(NEQS,1),GCB(1),A3(NEQ,1),XM(1),B(1),BO(1),VEL(1), 00022*16
*PDYL(2,4,1),TS(1),GACL(1),TD(2),HEDG(8),LD(NUMNP,1),R(1),A(1), 00023
*AO(1),Gb(1),PMAX(NUMEL,1),AMAX(NUMNP,12,2),DS(4,3),IH(NUMNP,1) 00024*25
                                00025**3
DIMENSION AS(NEQS,1)                                          00026**3
DIMENSION A4(NWA),A5(NEQ,1),A6(NWA),MAXA(1),X0(1),X1(1),X2(1),AT(00027
*1),BT(1),AC(1),SO(1),X(1),Y(1),Q(1),ISC(1),JSC(1),SURTRX(2,NDC,2),00028
*SURTRY(2,NDC,2),AN(NNSC,1),AM(1),EK(1),ANI(NYP,1),EKJ(NYP),NSLCI(1 00029**6
*),SURPFX(2,NDC,2),SURPFY(2,NDC,2)                            00030**2
                                00031**3
                                00032**3
DIMENSION RE(1)                                                00033*11
DIMENSION SW(1)                                                00034*15
COMMON/TAPES/NSTIF,NRED,NL,NR,IFIL(2),NUMBER,LCOUNT          00035*26
COMMON/PNTM/ISTRES,MPRINT,JPRINT,RPRINT,IF                    00036*25
                                00037
COMMON/EL/IND,LCOUNT                                           00038*18
                                00039*18
COMMON/BEWGI/SG(5,200,4,4),NCOND,NTEMS                        00040
REWIND 58
REWIND 59
                                00041*17
                                00042*17
                                00043*17
                                00044*17
                                00045*18
                                00046*17
                                00047*17
                                00048
****** TEMPORARY STORAGE ******
TR=NCYCL*DD                                                    00049
NC=0                                                            00050*23
DT=0.0                                                         00051*17
                                00052*17
                                00053*26
TM=0.0

```

455 CONTINUE

```

IND=NC
00054*20
00055*20
00056*10
00057*10
***** BRANCH FOR WORKING ARRAY INITIALIZATION IN ELT2D6 *00058*10
00059*10
00060*10
00061**5
00062**5
00063*20
00064*20
00065
00066
00067**5

REWIND 12

700 DO 1050 I=1,NEQ
1050 AU(I)=0.0

IF(NC.EQ.0) GO TO 701
LCOUNT=LCOUNT+1

701 CONTINUE
GO TO (400,300,400,400),KKS
300 IF(LOOP.EQ.1) GO TO 400
READ(12) (AU(I),I=1,NEQ)
GO TO 500

***** ASSEMBLE LOAD VECTOR***
400 CONTINUE

***** ELASTOPLASTIC ANALYSIS BRANCHING *****
IF(KKS.NE.3) GO TO 850
IF(NC.EQ.0) GO TO 850
CALL ELEMENT(NUMMAT)
CALL KSTAR(IH,A1,A3,XM,NEQ,MBAND,NEQH,MBANDH,A4,A5,A6,MAXA
*,NWA,NEQB,WBLOCK,M1,IB,KE,SW)

***** RETURN RESIDUAL STRESS VECTOR FOR SUBSEQUENT CALCULA
850 CONTINUE

IF(NDYN.EQ.2) GO TO 601

DO 880 I=1,NEQ
880 SU(I)=0.0

CALL LOAD1(NUMMT,NEQ,LC,ISLC,DI,NC,ISC,ISC,SURTRX,SURTRY,X,Y,000105

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```

*,ID,SU,IK,IM,PDYL,SURPFY,SURPFY)
00106**6
00107**6
00108**6
00109
00110
00111*10
00112*10
00113*10
***** MODIFIED LOAD VECTOR CALCULATION BOTH ELASTIC OR INE 00114*10
00115*10
00116*10
00117
9000 DO 9000 I=1,NEQ 00118**6
R(I)=0.0 00119**6
CALL LOADZ(NUMNP,PDYL,X,NUMLP,ID,DT,TD,NC,TK,IM,R) 00120**6
00121
8000 DO 8000 I=1,NEQ 00122
GB(I)=0.0 00123*16
00124*20
GO 10 600 00125*20
00126
00127*12
00128*12
***** EARTHQUAKE ANALYSIS ***** 00129*12
00130*13
00131*13
00132*12
601 CONTINUE 00133
CALL LOAD3(IS,GACL,DT,NACL,AM,ID,NUMNP,NC,TK,IM,GB,ACCL) 00134*16
600 CONTINUE 00135
00136
***** TOTAL LOAD VECTOR ASSEMBLED ***** 00137*12
00138*12
800 DO 800 I=1,NEQ 00139
AU(I)=SU(I)+R(I)+GB(I) 00140*16
00141*11
WRITE(6,46) (SU(I),I=1,NEQ)
WRITE(6,46) (AU(I),I=1,NEQ)
00142*22
00143*22
IF(KKS.NE.2) GO TO 500 00144**5
IF(LHUP.NE.1) GO TO 500 00145**5
WRITE(12) (AU(I),I=1,NEQ) 00146**5
500 CONTINUE 00147**5
00148**5
49 DO 49 I=1,NEQ 00149*20
GB(I)=AU(I) 00150*21
00151*25
00152*25
IF(KKS.NE.3) GO TO 1 00153*25
00154*25
KNORM=0.0 00155*20
50 DO 50 I=1,NEQ 00156*20
KNORM=KNORM+GB(I)*GB(I) 00157*21
00158*21
00159*23
2 CONTINUE 00160*23
00161*23
860 DO 860 I=1,NEQ 00162*23
AU(I)=AU(I)-RE(I) 00163*23

```

```

1      CONTINUE
      IF(NC.EQ.0) GO TO 10
      IF(LCOUNT.NE.1STRPR) GO TO 20
10     CONTINUE
45     WRITE(6,45)
      FORMAT(1H1/' ASSEMBLED LOAD VECTOR IN SOLV3 ***** '/')
46     WRITE(6,46) (GB(I),I=1,NEQ)
      FORMAT(5X,11E10.3)
20     CONTINUE
      IF(NC.NE.0) GO TO 46
      DO 43 I=1,NEQ
43      AO(I)=AO(I)-SW(I)
48     CONTINUE
      IF(NC.EQ.0) GO TO 42
***** CHECK OVERALL LOGIC FOR ALL MATERIAL MODEL **
      ISTRNS=ISTRNS+1
      MPRINT=MPRINT+1
      JPRINT=JPRINT+1
      KPRINT=KPRINT+1
42     CONTINUE
      IFLAG=0
      NITR=ITEMA
      ITR=0
      IF(NC.NE.0) GO TO 440
      CALL SYMSOL (1,A3,AO,NEQ,MBAND)
      CALL SYMSOL(2,A3,AO,NEQ,MBAND)
      IF(KKS.EQ.3) GO TO 451
      ***** KKS.EQ.3 AO(I) = INCREMENTAL *****
      ***** KKS.NE.3 AO(I) = TOTAL *****
      DO 450 I=1,NEQ
450  XU(I)=AO(I)+XU(I)
      GO TO 420
451     CONTINUE

```

```

00164*23
00165*23
00166*23
00167*27
00168*27
00169*27
00172*27
00173*27
00174*27
00175*27
00176*27
00177*20
00178*21
00179*21
00180*20
00181*21
00182*20
00183*27
00184*23
00185*23
00186*23
00187*23
00188*23
00189*20
00190**5
00191*12
00192*23
**00193*12
00194*12
00195**5
00196
00197
00198
00199
00200*17
00201*23
00202*20
00203*23
00204*23
00205*20
00206*20
00207*17
00208*17
00209*17
00210*17
00211*23
00212*22
00213*22
00214*22
00215*22
00216*22
00217*17
00218*19
00219*19
00220*23
00221*23
00222*23
00223*23
00224*23

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	IF(NEQUIB.NE.0) GO TO 453	00225*23
452	DO 452 I=1,NEQ	00226*23
453	XU(I)=XU(I)+AU(I)	00227*23
	CONTINUE	00228*23
		00229*19
440	GO TO 420	00230*17
	CONTINUE	00231*17
		00232*20
		00233*17
		00234*17
	SOLVE THE SIMULTANEOUS EQUATION FOR DISPLACEMENT VEC	00235
	CALL SULDN(AT,BT,XU,A5,NEQ,X1,X2,A4,NWA,AU,MAXA,NEQB,	00236*11
	*NBLOCK,MI,MBAND,NC ,XM)	00237
		00238*11
		00239
		00240
420	CONTINUE	00241*17
	IF(NEQUIB.EQ.0) GO TO 200	00242*17
		00243*17
		00244*17
	CALL DEQTR(XU,NDYN,AU,RE,R,ITR,NITR,IEMX,RTOL,NC,A4,NEQ,	00245*17
	*MBAND,NWA,NEQB,NBLOCK,MI,MAXA,X2,X1,VEL,AC,B,1D,AMAX,PMAX,AM,EK,AN	00246*19
	*,ANI,EKJ,IN,NUMNP,NUMEL,RNORM,XM,A5,GB,AT,BT,A3,LC)	00247*19
		00248*19
		00249*23
		00250*17
		00251*17
		00252
		00253
	INPUT DISPLACEMENT TO 'OUT' SUBROUTINE TO CALCUL	00254
	STRESS VECTOR AND RESPONSES *****	00255
		00256
200	CONTINUE	00257*10
		00258*10
		00259*17
	IFLAG=1	00260*17
	CALL OUT(X1,XU,X2,VEL,AC,AU,B,1D,DS,AMAX,PMAX,NUMNP,NUMEL,	00261*18
	*NC,AM,EK,AN,ANI,EKJ,TM,LC,NEQ,RE,ITR)	00262*18
		00263
		00264*11
		00265*23
		00266*23
		00267*17
		00268*17
	IF(IND.EQ.-1) GO TO 455	00269*18
		00270*18
		00271*18
	IF(NC.EQ.0) DT=DTA*DD	00272*24
	IF(NC.EQ.0) TM=TM+DT	00273*17
	IF(NC.EQ.0) GO TO 35	
	IF(LCOUNT.NE.1STRPR) GO TO 30	00274*27
	LCOUNT=0	00275*27
		00276*27
		00277*17
35	CONTINUE	

```
CALL FQCHR(XM,A5,B,NEG,MBAND,X0,X1,X2,R,NC,A3,NDYN,G5,A0,AS) 00278*24
30 CONTINUE 00279*23
    NC=NC+1 00280*27
    IF(NC-NC1EQ)700,700,900 00281**5
    900 CONTINUE 00282*22
    901 FORMAT(5X,315) 00283*22
    IF(KKS.NE.4) GO TO 990 00284**5
    IF(NDYN.NE.1) GO TO 990 00285*25
    IF(NTENS.NE.1) GO TO 990 00286
    ***** WRITE HERE LAST DISPL. AND VELOCITY COMPONENTS FOR REPEAT RUN
    WRITE(58,902) (X0(I),I=1,NEG)
    WRITE(59,902) (X1(I),I=1,NEG)
    902 FORMAT(5X,12E10.3)
    990 CONTINUE
    RETURN
    END 00287
    00288
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DATA SET SOLVED      AT LEVEL 035 AS OF 83/02/28
SUBROUTINE SOLVED(IH,A1,A3,XM,B,BO,VEL,PDYL,TS,GACL,TD,R,A,AU, 00001
                                00002*13
*GD,PMAX,AMAX,NEG,NUMNP,NACL,NUMEL,NEGH,GCB,A4,A5,A6,MAXA,X0,X1,X2,00003*21
                                00004*13
*AT,BT,AC,  SO,NWA,NEQB,NBLOCK,MI,X,Y,Q,ISC,JSC,SURTRX,SURTRY,AM,EK,00005
                                00006*13
*AN,NSLC1,ANI,EKJ,EC,SURPF1,SURPF2,RE,SW,AS) 00007*20
                                00008*13
IMPLICIT REAL*8      (A-H,O-Z) 00009
                                00010
COMMON/CONTR/DB,HEG(8),KADN,NASIM,NUMSL,NKK,MBAND,NCYCL,NDYN,IPLNA00011
*X,ISTRPK,NPRINT,ISTART,NEGK,MBANDH,NEQS,MBANDS 00012**3
COMMON/BOBBY/KKS,NUMMAT,NC,LOOP,NDRN,NINT,IDWA,NPT 00013*29
                                00014*29
COMMON/CONST/DELTA1,DELTA2,C1,C2,A0,A01,A02,A03,A04,A05,A06,A07,A08 00015
COMMON/INTEG /TETA,BETA,GAMA 00016
                                00017
                                00018
COMMON/PRIM/ISTRES,MPRINT,JPRINT,KPRINT,IP 00019*32
COMMON/SOL/ANORM,NGLOCK,NEDB,NDC
                                00020**3
                                00021**3
DIMENSION A1(NEQS,1),GCB(1),A3(NEG,1),XM(1),B(1),BO(1),VEL(1), 00022*21
*PDYL(2,4,1),TS(1),GACL(1),TD(2),HEG(8),ID(NUMNP,1),R(1),A(1), 00023
*AO(1),GD(1),PMAX(NUMEL,1),AMAX(NUMNP,1,2,2),DS(4,3),IH(NUMNP,1) 00024*32
                                00025**3
DIMENSION AS(NEQS,1)
DIMENSION A4(NWA),A5(NEG,1),A6(NWA),MAXA(1),X0(1),X1(1),X2(1),AT( 00026**3
*1),BT(1),AC(1),SO(1),X(1),Y(1),Q(1),ISC(1),JSC(1),SURTRX(2,NDC,2), 00027
*SURTRY(2,NDC,2),AN(NSC,1),AM(1),EK(1),ANI(NYP,1),EKJ(NYP),NSLC1( 00028
*1),SURPF1(2,NDC,2),SURPF2(2,NDC,2) 00029*11
                                00030**2
                                00031**3
                                00032**3
COMMON/ASIM/NYP,NSC,NSLC,NLC,NNYP,NNSC,ISLC 00033*17
COMMON /TAPES/ NBTIF,NREB,NI,NR,IFILL(2) 00034
COMMON/GRAND/COH(20),PHI(20),ITYPE(20) 00035
                                00036*32
                                00037*32
DIMENSION RE(1) 00038*14
DIMENSION SW(1) 00039*20
                                00040
                                00041*19
***** TEMPORARY ALLOCATION FOR SURFACE LOAD VECTOR 00042*19
COMMON/KRISNA/SCALB(50),SCALF(50),LL 00043*23
                                00044*29
                                00045*29
                                00046*23
                                00047
CALL      INIT(NUMEL,NUMNP,NEG,PMAX,AMAX,TD, X0,X1,X2,A,BO,AO,VEL 00048**6
*,B) 00049**6
                                00050
                                00051
                                00052
CALL CONST 00053
                                00054

```

LLL=NDYN

IF (LOOP.NE.1) GO TO 195

IF (KKS.EQ.4.AND.LC.GT.1) GO TO 195

*****READ HERE LINEARIZED YIELD SURFACE *****

DO 160 N=1,NUMMAT

READ(5,150) ITYPE(N)

160 READ(5,170) COH(N),PHI(N)

150 FORMAT(15)

170 FORMAT(2E10.3)

180 FORMAT(1H1/'***** TYPE OF LINEARIZED YIELD MODEL PROPOSED = ',I5/

* ' EQ. -1 NONE /

* ' EQ. 0 GIUDA METHOD /

* ' EQ. 1 ANDERHEGGEN MODEL /

* ' EQ. 2 MOHR COLUMB MODEL /)

WRITE(6,190) ((N,ITYPE(N),COH(N),PHI(N)),N=1,NUMMAT)

192 CONTINUE

190 FORMAT(/5X,' MATERIAL NUMBER = ',I5 /

* ' ITYPE(N) = ',I5/

* ' COHESIONCOH(N) = ',E10.3/

* ' ANGLE OF INTERNAL FRICTION = ',E10.3/)

195 CONTINUE

GO TO (550,550,550,580),KKS

580 CONTINUE

IF (LC.GT.1) GO TO 800

CALL YLDLIN(LC,NYP,NSC,ANI,EKJ)

550 CONTINUE

***** FOR ELASTO PLASTIC ANALYSIS READ MATERIAL PROPERTIES

00055

00056

00057*32

00058*32

00059*32

00060

00061

00062

00063*24

00064*25

00065*25

00066*32

00067*25

00068*24

00069*24

00070*24

00071*24

00072*24

00073*24

00074*24

00075*24

00076*24

00077*24

00078*24

00079*24

00080*33

00081*32

00082*31

00083*30

00084*24

00085*24

00086*24

00087*32

00088*24

00089*24

00090*24

00091*24

00092*24

00093

00094

00095**9

00096**9

00097**3

00098**3

00099*16

00100*16

00101**3

00102*16

00103*16

00104**3

00105*16

00106*16

00107**3

00108*10

00109*10

00110*12

00111*12

00112*12

00113*12

00114

IF(NDYN.EQ.2) GO TO 8000	00115
IF(LOOP.NE.1) GO TO 8000	00116
	00117*10
	00118**9
READ(5,505) ISLC,(NSLC1(K),K=1,NSLC)	00119**9
	00120*23
	00121*23
	00122*23
505 FORMAT(16I5)	00123*16
506 FORMAT(1H1//, ' INPUT TABLE 1B. MULTIPLE LOADING CASES ' //	00124*16
15X, 'LOADING', 10X, 'NUMBER OF SURFACE'	00125
25X, 'CASE NO.', 10X, 'LOAD CARDS' //)	00126
507 FORMAT(5X,15,19H . . . ,15,15)	00127*17
	00128
ISLC = 0 PROPORTIONAL CONCENTRATED LOADS	00129
ISLC = 1 VARIABLE REPEATED SURFACE TRACTIONS,	00130
AND PROPORTIONAL CONCENTRATED LOADS	00131
ISLC = 2 VARIABLE REPEATED SURFACE TRACTIONS,	00132
VARIABLE REPEATED CONCENTRATED LOADS,	00133
AND PROPORTIONAL CONCENTRATED LOADS	00134
	00135
	00136
	00137
	00138
800 CONTINUE	00139
	00140*16
	00141*16
NSLC=NSLC1(LC)	00142*17
WRITE(6,506)	00143*17
WRITE(6,507) LC,NSLC,ISLC	00144*17
8000 CONTINUE	00145
	00146
	00147
	00148
	00149**9
500 CONTINUE	00150**9
	00151**9
	00152*19
	00153*34
IF(LOOP.NE.1) GO TO 50	00154*34
	00155*19
LL=NSLC	00156*19
	00157*22
	00158*22
IF(LL.EQ.0) GO TO 49	00159*22
	00160*22
	00161*22
READ(5,45) (SCALB(I),I=1,NSLC)	00162*22
IF(NDRN.EQ.0) GO TO 50	00163*28
	00164*28
	00165*28
	00166*27
51 CONTINUE	00167*27
	00168*27
	00169*26
	00170*26
	00171*22
HEAD(5,45) (SCALF(I),I=1,NSLC)	00172*22
50 CONTINUE	00173*22
	00174*22

CALL	SOLV2 (IH,A1,A3,XM,B,BU,VEL,PDYL,TS,GACL,LD,R,A,AU,	00235
		00236*13
*GB,PMAX,AMAX,NEG,NUMNP,NACL,NUMEL,NEQH,GCB,A4,A5,A6,MAXA,XU,X1,X2,		00237*21
		00238*13
*AT,BT,AC, SU,HWA,NEQB,NBLUCK,M1,X,Y,Q,ISC,JSC,SURTRX,SURTRY,AM,EK,		00239
		00240*13
*AN,NSLC1,ANI,EKJ,LC,SURPFX,SURPFY,RE,SW,AS)		00241*20
		00242*13
GU TO 700		00243
		00244
		00245
		00246
700 RETURN		00247
END		00248

```

DATA SET STDSL0 AT LEVEL 051 AS OF 83/02/11
SUBROUTINE STDSL0(MF,LK,1H,LH,NUMNP,NEQ,FK,DENS,DENF,GMS,
00001*39
00002*39
00003*39
*VUL,K,F,EM,1MA1,KS,KF,THICK)
00004*26
IMPLICIT REAL*8(A-H,O-Z)
00005
00006*23
00007*23
COMMON/QUAD/RR(4),ZZ(4),FAC,R,Q(6),P(4,12),SF(12,12),H(12,12),
00008*39
*ST(20,12),D(4,4),C(12,12),E(12,12),IT(5,12),PI(4,8),UDD(4,4)
00009*25
COMMON/BELA/NS,NELTYP,ND,NUMGT,MODEL,NCUN,1DW,NK015,NK016,NK017
00010*39
00011*19
00012*48
COMMON/BOBBY/KKS,NUMMAT,NC,LOOP,NDRN,NINT,1DWA,NPT
00013*48
COMMON/ASHK/F1,F2,F3,AL,ALF,FACC,FACT
00014*25
COMMON/CONTE/DT,HEB(8),KADN,NASIM,NUMEL,NEQ,MBAND,NCYCL,NDYN,1PNAX
00015*25
*,1STRPK,NPRINT,1START,NEQH,MBANDH,NEQS,MBANDS
00016*51
00017*25
***** DIMENSION AND DATA FOR ROTATION OF STRESSES *****
00018*25
DIMENSION IVECT(4),JVECT(4),V(4)
00019*25
DATA IVECT/4,2,1,3/,JVECT/1,3,2,4/
00020*25
00021*25
00022*19
DIMENSION NP(4),LK(8),LH(8),1H(NUMNP,1)
00023*39
00024*19
00025*19
00026*19
DIMENSION SSS(5),FTT(5)
00027*19
DATA SSS/0.,-1.,1.,0.,0./,FTT/0.,0.,0.,-1.,1./
00028*55
DIMENSION GMS(8),FK(1),SK(8,8)
00029
00030*30
DIMENSION EM(1)
00031
00032*47
00033*47
00034
00035
00036*19
00037*19
00038*19
***** CALCULATE STRESS DISPLACEMENT MATRIX FOR ELEMENT
***** TOTAL STIFFNESS OF ELEMENT CONSIDERING UNDRAINED
00039*25
00040*25
00041*47
00042*47
00043*47
00044*47
00045*25
00046*25
00047*25
00048*25
00049*19
00050*23
00051*23
00052*23
00053*23
00054*23
160 WRITE(6,160)
FORMAT(5X,' ** PRINT IN STDSL0 **')
161 WRITE(6,161) NK015,NK016,NK017
FORMAT(5X,315)
350 CONTINUE
***** TRIANGLE STIFFNESS CONDITION AND STRESS OUTPUT REQUI
IF( (NF(3).EQ.NF(4)).AND.(ND.EQ.20)) ND=10

```



```

LL=NS/4
DU 530 L=1,LL
CALL FORMB(SSS(L),TIT(L),IPMAX,THICK)
***** FLUID DISPLACEMENT FUNCTION ARE CALCULATED ONLY FOR
IF(L.NE.1) GO TO 600
DU 900 I=1,4
DU 900 J=1,8
900 P1(I,J)=P(I,J)
***** CALCULATE FLUID STRAIN - DISPLACEMENT *****
500 CONTINUE
DU 600 JJ=1,ND
TT(L, JJ)=P(1, JJ)+P(2, JJ)+P(3, JJ)
600 CONTINUE
***** CALCULATE SOLID STRESS DISPLACEMENT MATRIX *****
DU 530 II=1,4
I=II+4*(L-1)
DU 530 J=1,ND
ST(I, J)=0.0
DU 530 K=1,4
***** DELETE TEMPORARILY FOR STRESS MATRIX CONDENSATION *****
ST(I, J)=ST(I, J)+D(II, K)*P(K, J)
530 CONTINUE
***** END OF STRESS DISPLACEMENT MATRIX *****
***** CONDITION FOR TRIANGULAR ELEMENT OF NON - INCOMPATIBLE
IF(NELTYP.EQ.0) GO TO 400
IF(NP(3).EQ.NP(4)) GO TO 400
***** CONDENSATION OF ST, H, ST, TT FROM 12 *12 TO 8
***** STATIC CONDENSATION OF INCOMPATIBLE NODES *****
DU 550 NN=1,4
L=12-NN

```

```

00055*23
00056*19
00057*39
00058*39
00059*36
00060*36
00061*36
00062*19
00063*33
00064*33
00065*33
00066*33
00067*33
00068*33
00069*33
00070*33
00071*19
00072*19
00073*19
00074*19
00075*19
00076*19
00077*19
00078*19
00079*19
00080*19
00081*19
00082*19
00083*19
00084*19
00085*19
00086*19
00087*19
00088*19
00089*19
00090*25
00091*47
00092*47
00093*47
00094*47
00095*25
00096*25
00097*25
00098*19
00099*19
00100*23
00101*23
00102*23
00103*23
00104*23
00105*23
00106*33
00107*33
00108*33
00109*23
00110*19
00111*19
00112*19
00113*19
00114*19

```

```

K=L+1
DU 550 J=1,L
GP=SF(I,K)/SF(K,K)
IF(NDRN.EQ.0) GO TO 560
560 CD=H(I,K)/H(I,K)
CONTINUE
DU 540 J=1,NS
GG=1.0
540 ST(I,J)=ST(I,J)-GP*ST(I,K)
***** CHECK CRITICALLY *****
***** FLUID VOLUMETRIC STRAIN COMPONENT CONDENSATION *****
DU 550 J=1,L
ST(I,J)=ST(I,J)-GP*SF(K,J)
IF(NDRN.EQ.0) GO TO 550
H(I,J)=H(I,J)-CD*H(K,J)
550 CONTINUE
***** END OF CONDENSATION *****
***** ROTATE AT THIS POINT STRESS-DISPLACEMENT MATRIX ACCO *****
***** TO S A P - 4 PROBLEM *****
400 CONTINUE
NSET=LL-1
IF(NSET.LE.0) GO TO 730
DU 720 L=1,NSET
IV=IVECT(L)
JV=JVECT(L)
CALL VECTPR(V,RR(IV),ZZ(IV),0.000,RR(JV),ZZ(JV),0.000)
S2=V(1)*V(1)
C2=V(2)*V(2)
SC=-V(1)*V(2)
I1=4*L+1
I2=I1+1
I4=I1+3
DU 710 J=1,8
B1=ST(I1,J)

```

```

00115*19
00116*19
00117*22
00118*42
00119*42
00120*42
00121*42
00122*19
00123*42
00124*19
00125*25
00126*22
00127*23
00128*23
00129*23
00130*23
00131*33
00132*33
00133*19
00134*22
00135*41
00136*41
00137*41
00138*41
00139*41
00140*41
00141*19
00142*19
00143*19
00144*19
00145*19
00146
00147*24
00148*24
00149*24
00150
00151
00152
00153*15
00154*15
00155*23
00156*23
00157*23
00158*25
00159*25
00160*25
00161*25
00162*25
00163*25
00164*25
00165*25
00166*25
00167*25
00168*25
00169*25
00170*25
00171*25
00172*25
00173*25
00174*25

```

B2=ST(12,J)	00175*25
B4=ST(14,J)	00176*25
B5=2.0*SC+B4	00177*25
ST(11,J)=C2*B1+S2*B2+B5	00178*45
ST(12,J)=S2*B1+C2*B2-B5	00179*25
710 ST(14,J)=SC*(B2-B1)+(C2-S2)*B4	00180*25
720 CONTINUE	00181*25
730 CONTINUE	00182*25
	00183*25
	00184*25
	00185*23
	00186*23
VLD=VOL/4.	00187*35
M=0	00188
DO 690 I=1,4	00189
IIM=1	00190
N=NP(IIM)	00191
DO 690 J1=1,2	00192
M=M+1	00193
LK(M)=LH(N,J1)	00194
690 CONTINUE	00195
	00196*50
GO TO (700,800,700,800),KRS	00197*50
	00198*50
800 CONTINUE	00199*50
	00200*50
DO 300 I=1,8	00201**6
GMS(I)=VLD*(DENS-(FK(K)**2.)*DENT)	00202*35
DO 300 J=1,8	00203**6
300 SK(1,J)=SF(1,J)	00204**6
	00205
	00206
	00207
700 CONTINUE	00208*15
	00209
K=IMAT	00210*37
F=FK(K)	00211**9
	00212*14
	00213*14
	00214*31
	00215*31
	00216
WRITE(51) (LK(I),I=1,8),(LH(I),I=1,8),((SK(I,J),J=1,8),I=1,8),	00217**5
*(PI(I,J),J=1,8),I=1,4),(GMS(I),I=1,8),VOL,DENS,DENT,F,NP	00218*12
	00219*13
	00220*13
*****THIS COMPLETES FOR WRITING ONE SET OF INFORMATION FOR ONE	00221
	00222
250 CONTINUE	00223*32
	00224
850 CONTINUE	00225*19
RETURN	00226
END	00227

DATA SET STIFF	AT LEVEL 012 AS OF 82/01/05	00001**9
SUBROUTINE STIFF(A1,NEQS,MBANDS,AS)		00002
IMPLICIT REAL*8(A-H,O-Z)		00003
		00004
		00005
DIMENSION A1(NEQS,1),AS(NEQS,NEQS)		00006**9
		00007**2
		00008**2
		00009**2
		00010
999 FORMAT(//)		00011**3
800 FORMAT(5X,RE12.5)		00012**3
DO 550 I=1,NEQS		00013
IF(I.NE.1) GO TO 530		00014
DO 590 J=1,MBANDS		00015**4
590 AS(I,J)=A1(I,J)		00016**2
GO TO 550		00017
530 JJ=I		00018
DO 560 J=1,MBANDS		00019**4
IF(JJ.GT.NEQS) GO TO 550		00020
AS(I,JJ)=A1(I,J)		00021**3
560 JJ=JJ+1		00022
550 CONTINUE		00023
DO 5930 I=1,NEQS		00024
DO 5930 J=1,NEQS		00025
AS(J,I)=AS(I,J)		00026
930 CONTINUE		00027
		00028
WRITE(0,5931)		
931 FORMAT(1H1/5X, ' PRINT OF STIFFNESS MATRIX '		
WRITE(0,5932) ((AS(I,J),J=1,NEQS),I=1,NEQS)		
932 FORMAT(5X,4E10.3)		00029
		00030
		00031
		00032
CALL INSYMA(NEQS,AS)		00033**9
		00034*11
WRITE(0,5931)		
WRITE(0,5932) ((AS(I,J),J=1,NEQS),I=1,NEQS)		
500 CONTINUE		00035*11
		00036
		00037
RETURN		00038
END		00039

DATA SET STIFF AT LEVEL 012 AS OF 82/01/05
 SUBROUTINE STIFF1(A1,NEQS,MBANDS,AS)
 IMPLICIT REAL*8(A-H,U-Z)

00001**9
 00002
 00003
 00004
 00005
 00006**9
 00007**2

DIMENSION A1(NEQS,1),AS(NEQS,NEQS)

551 DU 551 I=1,NEQS
 DU 551 J=1,NEQS
 AS(I,J)=0.0

00008**2
 00009**2
 00010
 00011**3
 00012**3
 00013
 00014
 00015**4
 00016**2
 00017
 00018
 00019**4
 00020
 00021**3
 00022
 00023
 00024
 00025
 00026
 00027
 00028
 00029
 00030
 00031
 00032
 00034*11
 00035*11
 00036
 00037
 00038
 00039

999 FORMAT(//)
 5800 FORMAT(5X,8E12.5)
 DU 550 I=1,NEQS
 IF(I.NE.1) GO TO 530
 DU 590 J=1,MBANDS
 590 AS(I,J)=A1(I,J)
 GO TO 550
 530 JJ=1
 DU 560 J=1,MBANDS
 IF(JJ.GT.NEQS) GO TO 550
 AS(I,JJ)=A1(I,J)
 560 JJ=JJ+1
 550 CONTINUE
 DU 5930 I=1,NEQS
 DU 5930 J=1,NEQS
 AS(J,I)=AS(I,J)
 5930 CONTINUE

8500 CONTINUE
 RETURN
 END

```

DATA SET STRAIN AT LEVEL 015 AS OF 83/02/24
SUBROUTINE STRAIN(PMAX,NUMEL,NUMMAT,GMX,DPL,GUSE,E,EN,PO)
IMPLICIT REAL*8(A-H,O-Z)
COMMON/POUT/NUMEL,NE(20)
COMMON/SOL/ANOKR,NBLOCK,NEQB,LE,NF,IDUM,NEIG,NAD,NVV,NFO
COMMON/BOBBY/KKS,NUGMAT,NC,LOOP,NDRN,NINT,IPWA,NPT

DIMENSION EITR(300),GITR(300),PULS(300),DAMP(300)

DIMENSION PMAX(NUMEL,1),GMX(1),DPL(1),GUSE(1),EN(1),E(1),PO(1)
DIMENSION LM(8),LH(8),SK(8,8),P(4,8),GMS(8),SF(8,8),EG(8,8),CC(8,8),
*) ,H(8,8),NP(4)
DIMENSION XP(16),XN(16)

REWIND 22
REWIND 51

KK=1

WRITE(6,21) LOOP

DO 80 M=1,NUMEL
  READ(51) ((LM(I),I=1,8),(LH(I),I=1,8),((SK(I,J),J=1,8),I=1,8),
  *) ((P(I,J),J=1,8),I=1,4),(GMS(I),I=1,8),VOL,DENS,DENF,F,NP

  READ(51) ((SF(I,J),J=1,8),I=1,8),((EG(I,J),J=1,8),I=1,8),((CC(I,J),
  *) J=1,8),I=1,8),LMAI,KS,KF,ALF,((H(I,J),J=1,8),I=1,8)
  *,(XP(I),I=1,16),(XN(I),I=1,16)

  K=INAT

  IF(LOOP.EQ.1) GO TO 90

  READ(22) ENEW,GNEW,DNEW,PSN

90 CONTINUE

81 CONTINUE

  IF(NAD.EQ.0) GO TO 300

  IF(M.EQ.NB(KG)) GO TO 100

300 CONTINUE

  IF(LOOP.NE.1) GO TO 95

```

```

00001**4
00002**2
00003**8
00004**8
00005*10
00006*10
00007*10
00008*15
00009**9
00010**8
00011**6
00012**3
00013**2
00014**4
00015**4
00016**6
00017**6
00018**2
00019**2
00020**2
00021**2
00022**7
00023*13
00024*13
00025*13
00026**2
00027**2
00028**8
00029**2
00030**2
00031**4
00032**2
00033**5
00034**5
00035**4
00036**2
00037**6
00038**2
00039**2
00040**7
00041**9
00042**9
00043**9
00044**9
00045**9
00046**9
00047**9
00048**7
00049**7
00050**8
00051**8
00052**8
00053**8
00054**8
00055**8
00056**8
00057**9
00058**9
00059**9

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	GNEW=GUSE(K)	00060**9
	DNEW=DPL(K)	00061**9
	AMAX1=PMAX(M,8)/GUSE(K)	00062*10
	GO TO 96	00063*10
		00064**2
		00065**9
		00066**9
		00067**9
95	CONTINUE	00068**9
	AMAX1=PMAX(M,8)/GNEW	00069**9
		00070**9
		00071**9
96	CONTINUE	00072**9
	AMAX1=100.*AMAX1	00073**9
	AJ=0.05*AMAX1	00074**2
	CALL CURV52(AJ,GN,DN)	00075**2
	GN=GN+GMX(K)	00076**2
	DN=DN/100.	00077**2
		00078**2
		00079**8
		00080**8
100	GO TO 200	00081**8
	CONTINUE	00082**8
	GN=GUSE(K)	00083**8
	DN=DPL(K)	00084**8
	GNEW=GN	00085*10
	DNEW=DN	00086*10
		00087*10
		00088**8
		00089**8
	KG=KG+1	00090**8
200	CONTINUE	00091**8
		00092**8
	DE=100.*(GNEW-GN)/GN	00093**9
	DZ=100.*(DNEW-DN)/DN	00094**9
	E1TR(M)=GN*Z.*(1.+PU(K))	00095**9
		00096**9
		00097**8
		00098**8
		00099*10
	WRITE(6,20) M,GNEW,GN,DNEW,DN,DE,DZ	00100*13
		00101*10
		00102*10
		00103*10
		00104**9
	G1TR(M)=GN	00105**9
	DAMP(M)=DN	00106**9
	POIS(M)=PU(K)	00107**9
		00108**9
		00109**7
		00110**7
		00111**7
80	CONTINUE	00112**2
	REWIND 22	00113*12
		00114*12
		00115**2
DO 99	M=1,NUMEL	00116*12
	WRITE(22) E1TR(M),G1TR(M),DAMP(M),POIS(M)	00117*10
99	CONTINUE	00118*10
		00119**4

```

00120**4
00121**4
65 FORMAT(// ' ELEMENT NO., USED SHEAR MODULUS, NEWLY CALCULATED SHEAR
1R MODULUS, PERCENTAGE DIFFERENCE ' //, 2X, 15, 6X, E17.9, 6X, E17.9, 10X, 2E15.9//)
00122
00123
00124
606 FORMAT(// ' ELEMENT NO., USED DAMPING VALUE, NEWLY CALCULATED VALUE
1, PERCENTAGE DIFFERENCE ' //, 2X, 15, 6X, E17.6, 6X, E17.6, 10X, E15.9//)
00125
00126
70 FORMAT( ' TEMPERATURE          E(N)          E(S)          E(T)          NU(NS)
1)  NU(ST)  NU(ST)          G(NS)  '/10X, 4E12.5, 3F10.4, E12.5//)
00127
00128
20  FORMAT(5X, 15, 6E12.5)
00129*13
00130*13
00131*13
21  FORMAT(1H1/ 40A, '***** ITERATION NO.          = ' 15//
+          40A, '          STRAIN COMP. SOIL PROPERTIES ' //
*5X, 'ELEM.' , 4X, 'GUSE' , 4X, 'GNEW' , 4X, 'DUSE' , 4X, 'DNEW' , 4X, 'G-DIFF' , 4X
* 'D-DIFF' //
*5X, ' NO. ' , 35A, '(PERCENT)' , 3X, '(PERCENT)' //)
00132*14
00133*14
00134*14
00135*14
00136*14
RETURN
00137
END
00138

```


	DATA SET STRCUM AT LEVEL 008 AS OF 83/02/18	00001
	SUBROUTINE STRCUM (NUMPAT,GMX,DPL,GUSE,E,EN,PU)	00002
	IMPLICIT REAL*8(A-H,I-Z)	00003**6
	COMMON/FOOT/NOEL,NL(20)	00004**6
	COMMON/SOL/ANOKM,NBLCK,NEQB,LL,NF,IDUM,NEIG,NAD,NVV,NFO	00005**6
	DIMENSION GMX(1),DPL(1),GUSE(1),E(1),EN(1),PU(1)	00006**6
	WRITE(6,101)	00007
101	FORMAT(1H1/5X,' PRINT IN STRCUM ')	
	REWIND 22	00008
	DO 30 N=1,NUMPAT	00009
	READ(5,100) GMX(N),DPL(N),GUSE(N),PU(N)	00010**2
	WRITE(6,102)GMX(N),DPL(N),GUSE(N),PU(N)	
102	FORMAT(5X,4E12.5)	
100	FORMAT(5X,4E12.5)	00011
30	CONTINUE	00012
	IF(NAD.EQ.0) GO TO 70	00013**6
	NAD=0 NO CONCRETE FOOTING	00014**6
	***** READ HERE CONCRETE FOOTING ELEMENT ****	00015**6
	READ(5,60) NOEL,(NL(1),I=1,NOEL)	00016**6
60	FORMAT(16I5)	00017**6
	WRITE(6,65) NOEL,(NL(1),I=1,NOEL)	00018**6
65	FORMAT(1H1/5X,' NO. OF FOOTING ELEMENT = ',15/	00019**6
*	5X,' FOOTING ELEMENT = ',6I5/)	00020**6
		00021**6
		00022**6
70	CONTINUE	00023**6
	DO 40 N=1,NUMPAT	00024**6
	E(N)=GUSE(N)*(2.*(1.+PU(N)))	00025**6
		00026**6
		00027
		00028**8
		00029**8
	EN(N)=E(N)	00030
40	CONTINUE	00031
50	CONTINUE	00032
	RETURN	00033
	END	00034

```
DATA SFT STRINV AT LEVEL 010 AS OF 83/01/15
SUBROUTINE STRINV(SIG, EPS, BLK, EMU, YOUNG, POIS, G, FKS, IPT, PF)
IMPLICIT REAL*8 (A-H, O-Z)
DIMENSION SIG(4), EPS(4)
DIMENSION PF(5)
DIMENSION G(4,4)
***** CALCULATE ELASTIC STRAIN BASED ON CURRENT DISPLAC
100 CONTINUE
DO 300 I=1,4
DO 300 J=1,4
300 SIG(I)=SIG(I)+G(I,J)*EPS(J)
DO 200 I=1,3
200 SIG(I)=SIG(I)+PF(I*PI)
RETURN
END
```

```
00001**7
00002**2
00003
00004**2
00005
00006**2
00007**6
00008**4
00009**2
00010**2
00011**4
00012**4
00013**4
00014**4
00015**4
00016*10
00017**4
00018**8
00019**8
00020**4
00021**2
00022
00023
```

```

DATA SET STSRPR      AT LEVEL 134 AS OF 83/03/05
SUBROUTINE STSRPR(B, TIME, PMAX, NUMEL, NNN, AM, EK, AN, ANI, EKJ, LC, ITR, 00001*94
*RE)                00002*94
                    00003*94
                    00004*33
                    00005*33
                    00006
                    00007
                    00008108
                    00009111
                    00010108
IMPLICIT REAL*8     (A-H,O-Z)
COMMON D(1)
COMMON/CONIL/DB, HED(8), KADN, NUMNT, NUMSI, NEG, MBAND, NCYCL, NDYN, IPLNA00011
*X, ISTRPR, WPRINT, ISTAR1, NEQH, MBANDH
*, NEUS, MBANDS
                    00012
                    00013118
                    00014*71
                    00015*71
DIMENSION ST(20,12), LM(16), SIG(9)
*, PS(16,8)
*, FACN(4)
                    00016*76
                    00017*56
                    00018*71
                    00019
DIMENSION G(4,4)
                    00020*82
                    00021
                    00022
COMMON/BELAY/MS, RELETP, ND, NUMGT, MODEL, NCON, IDW, NK015, NK016, NK017
COMMON/BOBY/KKS, NUMMAT, FC, LOOP, NDRN, NINT, IDWA, NPT
COMMON/NATMOD/STRESS(9), STRAIN(4), C(4,4), IPT, NEL
                    00023*32
                    00024*56
                    00025108
                    00026108
                    00027107
COMMON/SURIN/TI(5,12), XP(16), INDL, ITYP2D
COMMON/INTGE/TE TA, BETA, GAMMA, RTOL, NGAMA, NPUNCH, NEQUIB, ITEMX
                    00028104
                    00029*77
                    00030*32
***** BRING NLC HERE VIA COMMON *****
                    00031*44
COMMON/EL/IND, ICOUNT
                    00032*32
                    00033*35
                    00034*32
COMMON/GRK/ITLAC, IGR
COMMON/TAPES/NOO(6), NUMBER
                    00035115
                    00036130
                    00037130
                    00038115
***** TAKE IT TO ELASTIC PLASTIC ROUTINE TO UNDRAINED ****
COMMON/POBA/ALFA, EP, EPSV, SHRD
                    00039*96
                    00040134
                    00041134
                    00042*96
DIMENSION TO STORE DRESS RESPONSE FOR ELASTIC LIMIT LOAD
                    00043*96
                    00044*96
                    00045*96
COMMON/BENNY/SG(5,200,4,4), NCUND
                    00046117
                    00047*96
                    00048*96
                    00049*32
                    00050*32
***** PASS THIS VARIABLE FROM ELSTP IN DYNAMIC ALLOC
                    00051*32
                    00052*32
                    00053*32
                    00054108
                    00055*30
DIMENSION PE(5), P(4,8), ST(20,12), SIG(9), PS(16,8), FACN(4), G(4,4),
*LM(16)
DIMENSION B(1), PMAX(NUMEL,1)
                    00056
                    00057

```


	00107*44
	00108*44
***** FOR DYNAMIC LOADING STRESS VECTOR MAXIMIZATION ***	00109*44
	00110*44
IF(LC.GT.1.OR.NC.GT.1) GO TO 3	00111*44
	00112*46
CARD THDTH(NDC,NYP,NUMED,NED,NEY,AM,EM,AN,AMAX)	00113*17
	00114**2
	00115*44
	00116*44
	00117
3 CONTINUE	00118
RADZ=0.5*KADN	00119
PIZ=2.0 *DATA(1.000)	00120
	00121*39
	00122*69
IFEL=1	00123*69
	00124104
	00125127
NC=NNW	00126127
	00127127
	00128127
	00129104
IF(KKS.EQ.3.AND.FLAG.EQ.0) GO TO 87	00130114
	00131134
IF(NC.NE.0.AND.ISIKS.NE.1STRPR) GO TO 87	00132134
	00133134
IF(NPUNCH.NE.0) GO TO 2097	
	00134104
	00135104
WRITE(6,2020) TIME	00136*69
WRITE(6,2015) NC,ITR	00137*76
	00138*39
2097 CONTINUE	
IF(KKS.EQ.2.AND.LOOP.NE.NUMBER) GO TO 87	
WRITE(57,2099) TIME	
WRITE(10,2099) TIME	
2099 FORMAT(1X,' TIME STEP FOR STRESS OUTPUT =',F10.4)	
	00139114
87 IND=NNW	00140114
86 CONTINUE	00141*39
	00142113
	00143114
	00144114
	00145113
IK=0	00146*13
IP=1	00147*95
REWIND 10	00148*19
	00149*82
REWIND 50	00150*82
	00151*82
REWIND 17	00152*98
REWIND 22	00153128
	00154128
KK=1	00155*19
	00156*13
	00157*13
	00158*27

```

00159126
00160126
00161126
00162126
00163*27
00164*13
00165*13
***** TAPE 11 PASSED FROM ELSTIF ROUTINE
***** TAPE 13 PASSED FROM ELSTIF ROUTINE
00166*13
***** TAPE 12 CREATED IN THIS ROUTINE
00167*13
***** TAPE 17 CONTAINS INFORMATION ON DRAINED STRESS-STRAIN MATRIX
00168*98
00169*98
00170*13
00171*13
00172
00173*30
00174113
00175113
00176*30
00177*94
00178*94
00179*30
***** READ ELEMENT INFORMATION AS CREATED IN ELSTIF **00180*30
00181*30
00182*30
READ(11) ST,TT,P,IM,ALFA,EM,RM,ZM,KS,KF,IMAT,BLK,EMU,YOUNG,
*POIS,VOL,PS
00183*31
*,FACN
00184*56
00185*71
00186*30
SIMD=EMU
00187134
WRITE(6,103)NS,NELTYP,ND,NUMGT,MODEL,NCUN,LDW,NK015,NK016,NK017
00188134
00189*49
IF(KKS.NE.2) GO TO 345
00190128
IF(LOOP.EQ.1) GO TO 345
00191128
IF(NC.GT.0) GO TO 344
00192128
READ(22) LNEW,GNEW,DAMP,POIS
00193128
00194128
CALL MDCNEW(LNEW,GNEW,POIS,G)
00195128
GO TO 340
00196128
344 CONTINUE
00197128
00198128
IF(NEL.NE.1) GO TO 345
00199128
00200128
REWIND 17
00201128
00202128
00203128
00204128
00205128
00206129
00207129
00208129
00209128
00210128
00211128
00212128
345 CONTINUE
00213128
00214128
00215128
00216*86
00217*86

```

```

READ(17) ((G(I,J),J=1,4),I=1,4)
00218*87
00219*98
00220*86
00221*86
00222*86
340 CONTINUE
***** PUT A LOOP FOR CALCULATION OF STRESS OUTPUT *****
00223*87
00224*30
00225*30
PRINT 999
00226*47
00227*30
00228*99
301 DO 301 I=1,16
XP(I)=0.0
00229*99
00230*56
00231*56
00232*56
00233*56
00234101
LL=NPT
IF(NKS.EQ.4.AND.NCUND.NE.1) LL=NS/4
00235*56
00236*30
00237*30
00238*90
00239*30
DO 95 I=1,LL
FYID=0.0
00240*30
00241*30
00242*56
00243*56
00244*64
00245*64
PF(L)=0.0
IPT=L
WRITE(6,163)NS,NELTYP,ND,NUMGT,MODEI,NCUN,1DW,NK015,NK016,NK017
00246*64
00247*66
00248*66
00249
00250*19
00251*32
00252*10
00253
00254
00255*19
00256*19
00257*19
00258*19
00259117
00260117
00261*95
00262*33
00263*19
00264*19
00265*26
00266*26
00267*26
00268
00269
00270
00271123
00272123
00273121
DO 30 I=1,4
S1G(I)=0.0
30 STRAIN(I)=0.0
BB0=0.0
BB1=0.0
BB2=0.0
BB3=0.0
BB4=0.0
S=0.0
SP=0.0
EPSV=0.0
EPSVF=0.0
***** COMPUTATION OF SOLID STRAIN CALCULATION
DO 60 J=1,4
MM=1+J
MN=MM-1
DO 41 K=MN,MM

```

```

M=M+1 00274
***** 00275
IF(KS.EQ.0.AND.KF.EQ.0) GO TO 32 00276
IF(KS.EQ.1.AND.KF.EQ.-1) GO TO 32 00277
IF(KS.EQ.-1.AND.KF.EQ.1) GO TO 41 00278
***** 00279
32 N=LM(N) 00280
IF(N) 41,41,35 00281
35 CONTINUE 00282*30
00283*19
00284*19
00285*30
***** ELASTIC EFFECTIVE STRESS CALCULATION ***** 00286*30
***** IF(NDRN.NE.0) ... ST(I,J) EFFECTIVE STRESS DISPL. 00287*35
***** IF(NDRN.EQ.0) ST(I,J) ... TOTAL STRESS DISPLACE 00288*39
00289*39
00290*35
00291*35
00292*30
00293*30
00294*30
00295*30
00296*56
00297*94
00298*94
56 CONTINUE 00299*79
IF(NCOND.EQ.1) GO TO 55 00300*56
SIG(JJ)=SIG(JJ)+ST(J,K)*B(N) 00301*56
GO TO 40 00302*56
55 STRAIN(JJ)=STRAIN(JJ)+PS(J,K)*B(N) 00303*56
40 CONTINUE 00304*94
EPSV=EPSV+TT(1,K)*B(N) 00305*19
00306*19
00307
41 CONTINUE 00308*19
00309*19
00310*19
***** CALCULATION OF FLUID VOLUMETRIC STRAIN ***** 00311*19
00312*33
00313*33
00314*33
00315
00316
00317
00318
00319*38
00320*38
00321*38
00322*38
00323*38
00324*38
00325*19
00326*19
00327*26
42 EPSVF=EPSVF+TT(1,K)*B(N) 00328
00329
00330*56
00331*56
00331*56

```


PF(L)=EM*(CALPH+PEEMV+PEEMV)

00332*95

IF(KKS.EQ.4.AND.NC(0).NE.1) GO TO 57

WRITE(6,161)

FORMAT(7,1) PRINT IN STRSEP ** *)

WRITE(6,162) (STRAIN(I),I=1,4),PF(L)

WRITE(6,163) NC,ITF,IFLAG,NEH,L

WRITE(6,162) ((C(I,J),J=1,4),I=1,4)

00333*95

00334132

WRITE(6,163) IS,RELTYE,ND,NUMGI,MODEL,NCOR,TDW,NK015,NK016,NK017

WRITE(6,163) KKS,IC,PEE,L

00335132

00336*56

IF(KKS.EQ.3) GO TO 86

***** CALCULATE SOLID STRAIN INVERTING STRESSES *****

00337*57

00338*39

00339*39

00340*39

CALL STRINV(SIG,STRAIN,DIR,EMO,TH006,PO,6,KKS,ITF,PF)

00341*95

00342*49

00343*49

GO TO 59

00344*39

00345*39

00346*30

00347*56

88 CONTINUE

***** CALCULATE FLUID PRESSURE *****

00348*57

00349*30

00350*30

00351*30

WRITE(6,163) IS,RELTYE,ND,NUMGI,MODEL,NCOR,TDW,NK015,NK016,NK017

00352*94

57 CONTINUE

DO 58 I=1,3

SIG(I)=SIG(I)+PF(L)

00353*94

59 CONTINUE

00354*95

00355*93

IF(NPT.NE.4) GO TO 65

00356102

00357102

00358*96

00359*96

DO 60 I=1,4

SG(IC,NEH,L,I)=SIG(I)

00360*96

60 CONTINUE

00361*96

***** COMPLETION OF STRESS RESPONSE FOR ELASTIC LIMIT LOAD

00362*96

00363*96

00364*96

00365*19

00366*19

00367*27

00368*27


```

WRITE(6,163)IDW,NNN,NC,KKS,NFL,IP1
WRITE(6,163)NS,NELTYP,ND,NUMGT,MODEL,NCUN,LDW,NK015,NK016,NK017
WRITE(6,162) (D(I),I=NK015,NK016-1)
WRITE(6,162) (D(I),I=NK016,NK017-1)

```

```

CALL STSTN(LDW,PF,PS,EACH,IM,PMAX,
* NUMEL)

```

```

WRITE(6,163)NS,NELTYP,ND,NUMGT,MODEL,NCUN,LDW,NK015,NK016,NK017

```

```

00421*71
00422108
00425104
00423108
00424110
00426104

```

```

00427*53
00428*32
00429*32
00430*30
00431*30
00432*94
00433*94
00434*30

```

```

GO TO 95

```

```

**** EXCLUDING ELASTO PLASTIC ANALYSIS FOLLOW THESE STEPS *00435*30

```

```

00436*30
00437*19
00438*19
00439*19
00440*19
00441*19

```

```

160 CONTINUE

```

```

***** PRINCIPAL STRESS COMPUTATION *****00442*19

```

```

CC=(SIG(1)+SIG(2))/2.0
BB=(SIG(1)-SIG(2))/2.0
CR=DSQRT(BB**2+SIG(4)**2)
SIG(5)=CC+CR
SIG(6)=CC-CR
SIG(7)=(SIG(5)+SIG(6))/2.
SIG(8)=(SIG(5)-SIG(6))/2.

```

```

00443*19
00444*19
00445
00446
00447
00448
00449

```

```

IF (SIG(7).GT.10E-15.OR.BB.GT.10E-15) GO TO 80

```

```

SIG(9)=100.0

```

```

GO TO 90

```

```

80 SIG(9)=RAD2*DATAZ(SIG(4),BB)

```

```

90 CONTINUE

```

```

00450**7
00451**8
00452118
00453118
00454

```

```

00455**7
00456
00457**7
00458
00459
00460

```

```

COMPUTATION OF S1 & S2 FOR EACH TIME & FOR EACH SELECTED EL00462118

```

```

SM=(SIG(1)+SIG(2)+SIG(3))/3.
SX=SIG(1)-SM
SY=SIG(2)-SM
SZ=SIG(3)-SM
SXY=SIG(4)
SD=DSQRT(0.5*(SX*SX+SY*SY+SZ*SZ)+SXY*SXY)

```

```

00461*32
00463118
00464118
00465118
00466118
00467118
00468118
00469132

```

```

IF (STRAIN(4).EQ.0.0) GO TO 14

```

```

SHM=(SIG(4)/STRAIN(4))

```

```

00470131
00471131
00472131
00473134
00474134

```

```

14 CONTINUE
***** COMPUTE HERE STRESS COMPONENT P & Q ****
GO TO (12,12,12,13),NKS

13 CONTINUE
e IF(L.GT.1) GO TO 800
   KG=1MAT
   WRITE(6,799) NMMAT,1MAT,KG,KK
799 FORMAT(5X,415)
   IF(KK.NE.KG) GO TO 800
   READ(10)((ANI(J,1),1=1,NSC),J=1,NYP),(EKJ(J),J=1,NYP)
800 CONTINUE

   CALL TRDIA(NEL,NYP,NSC,ANI,AMD,IC,NDC,AM,EA,AN,NES,AMAX,EKJ,SIG,IK
*,NEY,IPT,VOL)

12 CONTINUE

   PG=PF(IPT)

   IF(NKS.EQ.4.AND.NCOND.NE.1) GO TO 52
   CALL ITRSD(NEL,I,SIG,FACN,PS,PT,PG,LM,XP)

52 CONTINUE

***** BRANCHING HERE FOR STRESS OUTPUT FOR DIFFERENT
IF(NC)808,805,808
808 IF(1STRES.NE.1STRES) GO TO 810

```

```

00475118
00476131
00477118
00478117
00479117
00480*32
00481**2
00482*33
00483*33
00484**2
00485*80
00486*80
00487*81
00488*80
00489*19
00490*19
00491*19
00492*19
00493*81
00494*81
00495*81
00496*13
00497*79
00498*33
00499*33
00500**3
00501*30
00592*95
00593*95
00590*96
00591*95
00589*96
00594100
00595100
00502*30
00503*95
00504*95
00505*30
00506*30
00507*90
00508*90
00509*30
00510*69
00511*69
00512*69
00513*69
00514*94
00515*94
00516*94
00517*94
00518*94
00519*69
00520*97
00521*97
00522*69

```

805 CONTINUE	00523*84
	00524*84
	00525*95
	00526*97
	00527*95
IF(L.NE.1) GO TO 820	00528*95
	00529*95
IF(NPUNCH.NE.0) GO TO 820	00530*95
	00531*84
	00532*84
WRITE(6,2003)	00533*84
WRITE(6,2004) NEL	00534*69
	00535*69
	00536*84
820 CONTINUE	00537*84
	00538*84
IF(NCOND.EQ.1) GO TO 821	
RM(L)=RM(L+NPT)	
ZM(L)=ZM(L+NPT)	
821 CONTINUE	
IF(NPUNCH.EQ.0)	
*WRITE(6,2007) L, RM(L), ZM(L), SIG, PF(IPT), FYLD	00539*90
IF(NKS.NE.2) GO TO 811	00540118
IF(LOOP.NE.NUMBER) GO TO 810	00541130
811 CONTINUE	00542118
	00543118
	00544128
IF(NPUNCH.EQ.0) GO TO 810	00545128
	00546128
	00547*90
	00548*97
	00549128
	00550128
IF(IP.GT.NEL) GO TO 810	00551128
	00552128
	00553128
LN=RELMT(IP)	00554*97
IF(LN.NE.NEL) GO TO 812	00555132
	00556*97
IF(L.NE.1) GO TO 814	00557*97
	00558102
	00559102
	00560128
	00561*97
	00562*93
	00563129
	00564129
	00565129

```

IF (STRAIN(4), EQ, 0.0) STOP=END
00566131
00567131
00568131
00569118
WRITE(57,203)EL, (SIG(1), I=1,2), SIG(4), (STRAIN(1), I=1,2), STRAIN(4)
00570119
*, SIG(5), SIG(6), SIG(8), SE, SD, F(CP1), SH, EPSV, FYID
00571119
203  FORMAT(13,7E11.4/0E10.3/)
00572119
00573119
814  CONTINUE
00574119
00575118
00576*97
IF=IF+1
00577102
00578*97
WRITE OTHER INFORMATION ON TAPE FOR PLOTTING
00579*97
00580*97
ALSO CALCULATE THE MAXIMUM STRESS STRAIN AND PURE PRESSURE
00581*97
00582*97
00583*93
00584*93
812  CONTINUE
00585*70
00587*95
00588*96
810  CONTINUE
00596*99
00597124
00598118
SIG(3)=SE
00599118
SIG(7)=SD
00600102
***** COMPUTE MAXIMA OF STRESSES AT EACH TIME STEP *****
00601102
00602102
DO 180 J=1,6
00603102
BB=DAFS(SIG(J))
00604103
IF (BB-PMAX(ME1,J)) 180,100,100
00605102
PMAX(ME1,J)=BB
00606102
PMAX(ME1,J+9)=TIME
00607102
180  CONTINUE
00608102
***** REDEFINE PURE PRESSURE *****
00609102
00610102
00611102
PP=DAFS(PG)
00612130
IF (PP-PMAX(ME1,9)) 200,200,200
00613130
PMAX(ME1,9)=PP
00614102
PMAX(ME1,10)=TIME
00615102
00616102
00617130
850  CONTINUE
00618130
00619130
IF (K5, NE, 2) GO TO 813
00620124
IF (LOOP, NE, NUMBER) GO TO 95
00621130
00622130
813  CONTINUE
00623124
00624128
IF (K) 800,809,800
00625128
00626128
00627128

```

```

806      CONTINUE
                                00628128
                                00629124
                                00630102
      ***** STRESS STRAIN CONTROL DATA *****
                                00631104
      IF(JPRINT.LI.ISTART) GO TO 95
                                00632104
                                00633104
                                00634128
809      CONTINUE
                                00635128
                                00636128
      IF(L.NF.1) GO TO 95
                                00637104
                                00638104
      ***** WRITE IT ON TAPE 18 *****
                                00639104
                                00640104
                                00641118
                                00642118
                                00643119
                                00644121
      *WRITE(18,204) NEL,SIG(1),SIG(2),SIG(4),SIG(8),PG,SHM,EPSV,STRAIN(400645132
      *)
      204 FORMAT(15,8E9.2,3X)
                                00646132
                                00647122
                                00648*95
                                00649*95
                                00650*69
                                00651*69
                                00652*35
                                00653*58
                                00654*75
      ***** WRITE ALL INFORMATION FOR OUTPUT ON TAPE *****
                                00655*75
                                00656*75
      ***** SELECT OPTION FOR STRESS OUTPUT *****
                                00657*75
                                00658*75
                                00659*75
                                00660*75
      IF(KRS.NE.4) GO TO 350
                                00661*95
                                00662*75
                                00663*75
                                00664*75
                                00665*75
                                00666*75
      IF(KK.G1.NUMMAT) REWIND 10
      IF(KK.G1.NUMMAT) KK=1
                                00667*75
                                00668*58
                                00669*95
                                00670102
                                00671116
                                00672116
                                00673116
                                00674104
                                00675115
                                00676115
                                00677115
                                00678132
      WRITE(6,162) (XP(I),I=1,16)
      WRITE(6,163) (LM(I),I=1,16)
162     FORMAT(5X,8E10.3)
163     FORMAT(5X,1615)

```

```
IF (KKS.FO.4.AND.NCOND.NE.1) GO TO 300
```

```
00679115
00680*99
00681*99
```

```
WRITE (21) LM, XF
WRITE (6,162) (XP(1), I=1, 16)
WRITE (6,163) (LM(1), I=1, 16)
```

```
300 CONTINUE
```

```
00682*70
00683*70
00684*70
00685131
```

```
IF (IFLAG.EQ.0) GO TO 311
IF (JPRINT.LI.ISTART) GO TO 311
JPRINT=0
```

```
00686131
00687131
00688104
00689104
```

```
311 CONTINUE
```

```
00690105
00691105
00692*71
00693131
```

```
IF (IFLAG.EQ.0) GO TO 310
```

```
00694131
00695*71
00696*71
00697*71
```

```
IF (ISTRES.LI.ISTKPR) GO TO 310
ISTRES=0
310 CONTINUE
```

```
00698*69
00699*69
00700*21
00701*13
```

```
IF (KKS.NE.4) GO TO 900
```

```
IF (LC.NE.NBC) GO TO 900
```

```
00702*44
00703*24
00704*24
00705*24
```

```
999 FORMAT(//)
```

```
00706
00707
00708
00709*44
```

```
PRINT 999
WRITE (6,6000)
6000 FORMAT(1H1/' **** MAX. LOAD VECTOR NORMALIZED FOR ALL LOAD
```

```
* CASES **** ')
WRITE (6,6001) (AM(J), J=1, NNYP)
PRINT 999
```

```
00710*44
00711
00712
00713
```

```
WRITE (6,6002)
6002 FORMAT(1H1/' **** K VECTOR ****')
WRITE (6,6001) (EK(J), J=1, NNYP)
```

```
00714*44
00715
00716
```

```
6001 FORMAT(5X,1ZF10.3)
```

```
900 CONTINUE
```

```
00717*19
00718*19
00719*21
00720*75
```

```
2015 FORMAT(//// 37H EQUILIBRIUM ITERATION IN TIME STEP = ,15//
* 37H NUMBER OF ITERATIONS = ,15 //)
```

```
00721*75
00723*76
00724*76
00725*76
00726*76
```

```
2020 FORMAT(1H1/' STRESSES AND FLOWS AT TIME = ',F10.3//)
```

```
00727*36
00728
00729
```

```
2001 FORMAT(1H1/' STRESSES AND FLOWS AT TIME = ',F10.3//')
* ELEMENT/' NUMBER',3X,'SIGR',5X,'SIGZ',6X,'SIGT',5X,'SIGRZ',5X,'S00730
*1GMAX',4X,'SIGMIN',5X,'MEAN-NORMAL',5X,'MAX-DEVIATION',5X,'ANGLE',
*5X,'PE1',4X,'PE2',4X,'PE3'//)
```

```
00731**7
00732*19
00733**7
```

```
2005 FORMAT(16,1ZF10.3)
```

```
2002 FORMAT(1H1/'***** STRAINS AND VOLUMES AT TIME *****= ',F10.3/00734**7
*// ELEMENT/' NUMBER',3X,'EPSVS',5X,'EPSVI',6X,'BULK-STRAIN',6X,'S00735**7
```



```

DATA SET S1S1H      AT LEVEL 012 AS OF 83/02/11
SUBROUTINE S1S1H(LUW,PF,PS,FACN,LM,PMAX,NUMEL)
IMPLICIT REAL*8(A-H,O-Z)
00001**9
00002
00003
00004
00005
00006
00007
SUBROUTINE
00008
TO FIND STRESS STRAIN LAW AND STRESSES FOR
NONLINEAR MATERIAL MODELS
00009
00010
00011
00012
00013
00014
00015
00016
COMMON/EL/IND,ICOUNT
00017**2
COMMON/VAR/NG,KPF1
00018**2
COMMON/GB/GA/NS,NELTYP,ND,NUMGT,MODEL,NCUN,IGW,NK015,NK016,NK017
00019**5
COMMON/HORBY/KKS,HUMMAI,NC,LDUP,NDRN,NINT,IDWA,NPT
00020*11
00021*11
00022*11
COMMON/SURIN/IT(5,12),XF(16),INDMB,ITYPZD
00023*12
COMMON/MATMOD/STRESS(9),STRAIN(4),C(4,4),IPT,NEL
00024**9
00025**9
DIMENSION PMAX(NUMEL,1)
00026**9
00027**9
00028**9
00029
DIMENSION PROP(NCUN),WA(IDWA,1)
00030*11
00031*11
00032*11
00033*10
DIMENSION PS(16,8),FACN(4),LM(16)
00034**8
00035
00036
DIMENSION PF(1)
00037**7
IST=3
00038
IF (ITYPZD.EQ.0) IST=4
00039
IP=NCUN/4
00040

WRITE(6,100)
100  FORMAT(5X,' *** PRINT IN S1S1H *** ')
WRITE(6,101)NS,NELTYP,ND,NUMGT,MODEL,NCUN,IGW,NK015,NK016,NK017
101  FORMAT(5X,10I5)
00041

DEFINITION OF STRAIN
00042
00043
00044
00045
LINEAR STRAIN TERMS
00046
00047
00048
00049
00050
CALCULATION OF STRESS-STRAIN
00051
MATRIX AND STRESSES
00052
00053
00054

```

```
80 GO TO (1,2),MODEL
```

00055**2

00056

00057

00058

00059

00060

00061

00062

00063**9

00064**9

00065**11

00066**12

00067**11

00068**11

00069**9

00070**9

```
1 CONTINUE
```

```
CALL ELI2D6(PF,FACN,PS,LM,PMAX,NUMEL)
```

```
WRITE(6,160)
```

```
WRITE(6,161) NS,NELTYP,ND,NUMGT,MODEL,NCON,IGW,NK015,NK016,NK017
```

```
RETURN
```

00071

00072

00073

00074

00075

00076

00077**9

00078**9

00079**9

00080**9

00081**8

00082**8

00083

00084

00085

00086

00087

00088

```
.... MODEL = 7 ELASTOPLASTIC (DRUCKER-PRAGER)
```

```
2 CONTINUE
```

```
CALL ELI2D7(PF,FACN,PS,LM,PMAX,NUMEL)
```

```
RETURN
```

```
END
```

DATA SET SYMSUB AT LEVEL 001 AS OF 81/04/28
 SUBROUTINE SYMSUB (KKK,A,B,NN,MM)
 IMPLICIT REAL * 8 (A-H,O-Z)

		00001
		00002
		00003
		00004
		00005
		00006
		00007
		00008
		00009
		00010
		00011
		00012
		00013
		00014
		00015
		00016
		00017
		00018
		00019
		00020
		00021
		00022
		00023
		00024
		00025
		00026
		00027
		00028
		00029
		00030
		00031
		00032
		00033
		00034
		00035
		00036
		00037
		00038
		00039
		00040
		00041
		00042
		00043
		00044

```

DIMENSION A(NN,1),B(1)

1  FORMAT (1015)
   GO TO (5,275),KKK
5  CONTINUE
5  DO 270 N = 1, NN
   DO 270 L = 1, MM
   DO 260 K = 2, MM
   IF(A(N,1).EQ.0) WRITE(6,1000) N,L,K
1000 FORMAT(5X,415)
   C = A(N,L)/A(N,1)
   I = N+L-1
   IF(NN-I) 260,240,240
240  J = 0
   DO 250 K = L, MM
   J = J + 1
250  A(I,J) = A(I,J)-C*A(N,K)
260  A(N,L) = C
270  CONTINUE
   RETURN

275  DO 290 N = 1, NN
   Y = B(N)
   DO 280 L = 2, MM
   I = N+L-1
   IF(NN-I) 290,280,280
280  B(I) = B(I) - A(N,I)+Y
290  B(N) = Y /A(N,1)

   N = NN
300  N = N - 1
   IF(N) 350,500,350
350  DO 400 K = 2, MM
   I = N+K-1
   IF(NN-I) 400,370,370
370  B(N) = B(N) - A(N,K)*B(I)
400  CONTINUE
   GO TO 300
500  RETURN
   END

```

```

DATA SET TRIFAC      AT LEVEL 003 AS OF 83/02/24
SUBROUTINE TRIFAC (A,B,MAXA,NEQB,NA,NBLOCK,NWA,NIB,NEQ,M1)
IMPLICIT REAL*(A-H,O-Z)
      1
CALLED BY:  STEP
THIS ROUTINE DECOMPOSES THE SYSTEM MATRIX IN BLOCKS
DIMENSION  A(NWA),B(NWA),MAXA(M1)
COMMON/TAPES/NSTIF,NEED,NL,NR,IFIL(2),NUMBER
      2
MA2=MA - 2
IF(MA2.EQ.0) MA2 = 1
INC=NEQB - 1
SET TAPE ASSIGNMENTS
NSTIF = 4
NREB = 3
NL = 1
NR=8
      3
N1=NL
N2=NR
REWIND NSTIF
REWIND NREB
REWIND N1
REWIND N2
MAIN LOOP OVER ALL BLOCKS
DO 100 NJ=1,NBLOCK
IF (NJ.EQ.1) GO TO 10
READ (NSTIF) A
GO TO 100
10 IF (NIB.EQ.1) GO TO 100
REWIND N1
REWIND N2
READ (N1) A
      4
FIND COLUMN HEIGHTS
KU=1
KM=MINO(MA,NEQB)
MAXA(1)=1
DO 110 N=2,M1
IF (N-MA) 120,120,130
120 KU=KU + NEQB
KK=KU
MM = MINO(N,KM)
GO TO 140
130 KU=KU + 1
KK=KU
IF (N-NEQB) 140,140,130
136 MM=MM - 1
140 DO 160 K=1,MM
IF (A(KK)) 110,160,110
160 KK=KK - INC
110 MAXA(N)=KK
IF(A(1)) 174,174,170
174 KK = (NJ-1)*NEQB + 1

```

```

00001
00002
00003
00004
00005
00006
00007
00008
00009
00010**3
00011
00012
00013
00014
00015
00016
00017
00018
00019
00020
00021**2
00022
00023
00024
00025
00026
00027
00028
00029
00030
00031
00032
00033
00034
00035
00036
00037
00038
00039
00040
00041
00042
00043
00044
00045
00046
00047
00048
00049
00050
00051
00052
00053
00054
00055
00056
00057
00058
00059

```

	IF (KK.GI.NEG) GO TO 590	00060
	WRITE (6,1000) KK	00061
	STOP	00062
172	KK = (ND-1)*NEQB + 1	00063
	WRITE (6,1010) KK	00064
		00065
	FACTORIZE LEADING BLOCK	00066
176	DO 200 N=2,NEQB	00067
	NH=MAXA(N)	00068
	IF (NH=N) 200,200,210	00069
210	NI=N + INC	00070
	NH=NH	00071
	K=N	00072
	D=0.	00073
	DO 220 KK=KL,KU,INC	00074
	K=K - 1	00075
	C=A(KK)/A(N)	00076
	D=D + C*A(KK)	00077
220	A(KK)=C	00078
	A(N)=A(N) - D	00079
		00080
	IF (A(N)) 222,224,230	00081
224	KK=(ND-1)*NEQB + N	00082
	IF (KK.GI.NEG) GO TO 590	00083
	WRITE (6,1000) KK	00084
	STOP	00085
222	KK = (ND-1)*NEQB + N	00086
	WRITE (6,1010) KK	00087
		00088
230	IC=NEQB	00089
	DO 240 J=1,MA2	00090
	ND=MAXA(N+J) - IC	00091
	IF (ND=N) 240,240,280	00092
280	KU=MINO(MJ,NH)	00093
	KH=N + IC	00094
	C=0.	00095
	DO 300 KK=KL,KU,INC	00096
300	C=C + A(KK)*A(KK+IC)	00097
	A(KU)=A(KU) - C	00098
240	IC=IC + NEQB	00099
		00100
200	CONTINUE	00101
		00102
	CARRY OVER INTO TRAILING BLOCKS	00103
320	DO 400 MK=1,MI	00104
	IF ((MK+ND).GI.NB[LUCK]) GO TO 400	00105
	NI=NI	00106
	IF ((ND.GI.1).OR.(MK.FI.NIB)) NI=ND+1	00107
	READ (MI) B	00108
	ML=MK*NEQB + 1	00109
	MR=MINO(MK+1)*NEQB,MI	00110
	ND = MI - MD	00111
	KL=NEQB + (MK-1)*NEQB+NEQB	00112
	N=1	00113
		00114
		00115
	DO 500 M=MD,MR	00116
	NH=MAXA(M)	00116
	KL=KL + NEQB	00117
	IF (NH-KL) 505,510,510	00118
510	KU=NH	00119

DATA SET VECTOR AT LEVEL 003 AS OF 81/11/19

SUBROUTINE VFCUR(V,XI,YI,ZI,XJ,YJ,ZJ)

IMPLICIT REAL* 8 (A-H,O-Z)

DIMENSION V(4)

X=XJ-XI

Y=YJ-YI

Z=ZJ-ZI

V(4)=DSQR(X**2+Y**2+Z**2)

V(3)=Z/V(4)

V(2)=Y/V(4)

V(1)=X/V(4)

RETURN

END

00001
00002**2
00003**3
00004
00005
00006
00007
00008
00009
00010
00011
00012

DATA SET YLDCHK AT LEVEL 010 AS OF 85/03/02

	SUBROUTINE YLDCHK(FYLD,IPMAX,SIG,IG,NOMEL,NPT)	00001*15
	IMPLICIT REAL * 8(A-H,O-Z)	00002
	DIMENSION SIG(9)	00003
		00004
		00005
	DIMENSION ALPHA(5,200,4),DF(20,12),DM(16),PS(16,8),FACN(4)	00006**6
	DI-FACN(5),ZM(5)	00007*15
		00008**6
	COMMON/BELOY/SG(5,200,4,4),NCOND	00009*13
		00010*13
	COMMON/GRAND/COH(20),PHI(20),TYPE(20)	00011*15
		00012*13
		00013*12
	COMMON/ANIM/NIT,NBC,NDFC,NDC,NMTP,NMBC	00014*12
		00015*12
	DIMENSION P(4,8),IT(5,12)	00016*12
		00017*12
		00018
	NDF=1	
	IF (NCOND.EQ.1) NDF=4	
900	WRITE(6,500) IPMAX,IG,NOMEL,NIT,NDC	00019
	FORMAT(16F5)	00020*16
		00021*16
		00022**6
		00023*12
	DO 100 I=1,NDC	00024*12
	DO 320 N=1,NOMEL	00025*12
	DO 320 I=1,NDF	00026**6
320	ALPHA(IC,N,I)=0.0	
	REWIND 11	00027**6
		00028**6
	DO 200 N=1,NOMEL	00029**7
		00030**6
	READ(11) SI,II,P,DM,ALPHA,FAC,DM,ZM,KD,KE,IMAT,BLK,EMU,YOUNG,P	00031**6
	*OIS,VOL,PS,FACN	00032**6
		00033**6
		00034**6
		00035**6
	DO 300 I=1,NDF	00036**6
		00037**6
400	DO 400 I=1,4	00038**6
	SIG(I)=SG(IC,N,I,I)	00039**6
		00040**6
		00041**6
	IF (IPMAX) GO TO 700,700	00042**6
		00043**6
600	SIGYZ=SIG(1)**2	00044**6
	SIGYZ=SIG(2)**2	00045**6
	SIGXY=SIG(1)*SIG(2)	00046**6
	SIGXYZ=3.*(SIG(4)**2)	00047**6
	Z=(SIGYZ+SIGYZ-SIGXY+SIGXYZ)	00048**6
	IT=COH(IMAT)	00049*18


```

XLD=YLD**2
IF(Z.FU.O.U) GO TO 751
ALPHA(C,N,L)=DSORT(YLD/Z)
COWT=TIME
WRITE(0,1000) LC,N,L,SIGX2,SIGY2,SIGAY,Z,YLD,CUH(IMAT),
ALPHA(C,N,L)
GO TO 300

```

751

*

```

CONTINUE
00050*18
00051*18
00052*18
00053*18
00054**6
00055**6
00056**6

```

700

```

IF(I TYPE(IMAT),FU.O) GO TO 702
IF(I TYPE(IMAT),FU.J) GO TO 702
IF(I TYPE(IMAT),EU.2) GO TO 703
SIGX=SIG(1)
SIGY=SIG(2)
SIGAY=SIG(4)
IF=IP*(DSIN(PHI(IMAT)))
YN=(SIGX-SIGY)
YN=YN**2
SAY=4.*(SIGAY)**2
TERM=(YNSAY)
TUD=(Z.*(CUH(IMAT))+(DCOS(PHI(IMAT))))+IP)
FLD=DSUR1(TERM)

```

703

```

GO TO 704
DM=(SIG(1)+SIG(2)+SIG(3))/3.
DMSIG(1)=DM
DY=SIG(2)-DM
DZ=SIG(3)-DM
DS=SIG(4)
A=3.*(CUH(IMAT)**2)+(DCOS(PHI(IMAT)))**2
B=3.*(DSIN(PHI(IMAT)))**2
C=(DSIN(PHI(IMAT)))**2
AL=(C/D)**(1/3.)
AL=DSUR1(AL)

```

702

```

SK=A/H
FLD=DSUR1(FLD)+AL*DM
IF(FLD.EQ.0.0) GO TO 750
ALPHA(C,N,L)=(YLD/FLD)

```

704

```

CONTINUE
WRITE(0,1000) LC,N,L,SIG(1),SIG(2),SIG(3),CUH(IMAT),PHI(IMAT)
*,YLD,TUD,ALPHA(C,N,L)
00068*16
00069*16
00070*16
00071*16
00072*16
00073*16
00074**6
00075**6
00076**6
00077**6
00078**6

```

750

```

1000 FORMAT(5X,3I5,HE10.3)

```

300

200

CONTINUE
CONTINUE

DATA SET YLDIN AT LEVEL 010 AS OF 82/06/12
 SUBROUTINE YLDIN(NSC,NYP,NUMEL,NES,NEY,AM,EK,AN,AMAX)

IMPLICIT REAL*8(A-H,O-Z)

COMMON/BOBBY/NEK,NUMG1,NC,LOOP,NDRN,NINT,LDWA,NPT
 DIMENSION AM(1),EK(1),AN(NES,1),AMAX(NYP,1,1)

COMMON/BENDY/SG(5,200,4,4),NCUND

NDT=1
 IF(NCUND.EQ.1) NDT=4

DO 2 I=1,NEY
 AM(I)=0.0
 2 EK(I)=0.0
 DO 4 I=1,NES
 DO 3 J=1,NPT
 3 AN(I,J)=0.0
 4 CONTINUE
 15 FORMAT(5X,12E10.3)
 DO 7 M=1,NUMEL
 DO 7 I=1,NDI
 DO 7 J=1,NYP

AMAX(J,M,1)=0.0

7 CONTINUE

RETURN
 END

00001
 00002**8
 00003
 00004**8
 00005*10
 00006*10
 00007*10
 00008**8

00009**8
 00010
 00011
 00012**3
 00013**4
 00014**3
 00015**7
 00016**4
 00017**6
 00018
 00019*10
 00020*10
 00021*10
 00022
 00023*10
 00024*10
 00025*10
 00026*10
 00027*10
 00028
 00029
 00030

```

DATA SET YDDIIN AT LEVEL 027 AS OF 83/02/28
SUBROUTINE YDDIIN(LC,NYP,NSC,ANI,EKJ)
IMPLICIT REAL*8(A-H,O-Z)
DIMENSION ANI(NYP,1),EKJ(NYP)
COMMON/CONTI/DB,REI(8),RADN,NUMNIP,NUMDI,NEQ,MBAND,NCYCL,NDYN,IPLNA
*X,ISTRK,MPRINT,ISTAKI,NEQH,MBANDH
*,NEQS,FBANDS
COMMON/BOBBY/KKS,NUMMAT,NC,LOOP,NDRN,NINT,IDWA,NPT
COMMON/GRAND/COH(20),PHI(20),ITYP(20)
COMMON/BENDY/BO(5,200,*,*),NCOND,NTENS
COMMON/GORA/IFLAG,ITR
DIMENSION ANG(14,14)
301 WRITE(6,301) NUMMAT,COH(1),PHI(1),ITYP(1)
FORMAT(5X,15,2E10.3,15)
PI=4.0*DATAH(1.000)
RADN=PI/180.
500 FORMAT(15)
400 FORMAT(1H1/' TYPE OF LINEARIZATION MODEL PROPOSED = ',I5/
* ' EQ.0. .... GIUDA METHOD ..... / /
* ' EQ.1 ..... ANDERHEGGEN METHOD / /
* ' EQ.2 ..... MONK CUBDOMB / /
200 FORMAT(5X,2E10.3)
REWIND 10
DO 300 N=1,NUMMAT
***** INITIALIZE *****
DO 700 J=1,NYP
DO 700 I=1,NYP
ANI(J,I)=0.0
ANG(J,I)=0.0
700 CONTINUE
WRITE(6,400) ITYP(N)
ITYPE=ITYP(N)
WRITE(6,200) COH(N),PHI(N)
PHI(N)=PHI(N)*RADN
WRITE(6,200) COH(N),PHI(N)

```

```

00001**6
00002**4
00003
00004**4
00005**4
00006
00007**2
00008**2
00009*24
00010*24
00011*24
00012**2
00013*26
00014*26
00015*25
00016*21
00017*24
00018*24
00019**2
00020**2
00021*14
00022**2
00023*27
00024*27
00025*16
00026*16
00027*16
00028*16
00029*16
00030**5
00031**5
00032**5
00033*16
00034*16
00035**2
00036**2
00037*11
00038**2
00039**2
00040**2
00041**2
00042**2
00043**2
00044**2
00045**2
00046**2
00047**2
00048*12
00049**7
00050**7
00051**7
00052*26
00053*26
00054*12
00055**2
00056**8
00054*12

```

```

0005/**8
0006/**8
0007/**8
0008/**8
0009/**8
0010/**8
0011/**8
0012/**8
0013/**8
0014/**8
0015/**8
0016/**8
0017/**8
0018/**8
0019/**8
0020/**8
0021/**8
0022/**8
0023/**8
0024/**8
0025/**8
0026/**8
0027/**8
0028/**8
0029/**8
0030/**8
0031/**8
0032/**8
0033/**8
0034/**8
0035/**8
0036/**8
0037/**8
0038/**8
0039/**8
0040/**8
0041/**8
0042/**8
0043/**8
0044/**8
0045/**8
0046/**8
0047/**8
0048/**8
0049/**8
0050/**8
0051/**8
0052/**8
0053/**8
0054/**8
0055/**8
0056/**8
0057/**8
0058/**8
0059/**8
0060/**8
0061**2
0062**2
0063**3
0064**2
0065**3
0066**2
0067**2
0068**2
0069**2
0070**2
0071**2
0072**2
0073**2
0074**2
0075**2
0076**2
0077**2
0078**2
0079**2
0080**2
0081**2
0082**5
0083**5
0084**5
0085**5
0086**6
0087*16
0088**5
0089**5
0090**5
0091*16
0092*16
0093**5
0094**5
0095**5
0096**5
0097**5
0098**15
0099**5
0100**5
0101**5
0102**5
0103**5
0104**5
0105**5
0106**5
0107**5
0108**5
0109**5
0110**5
0111**5
0112**5
0113**5
0114**5
0115**7
0116**7

```

650 IF (PII(1)) GO TO 800
CONTINUE
IF (I(1,1)) GO TO 800

AGM=(1.+DSIN(PII(1)))/(1.-DSIN(PII(1)))
AA=DSQR(1.+AGM**2.)
BB=(2.*COS(PII(1))*COS(PII(1)))/(1.-DSIN(PII(1)))
ANI(1,1)=1./AA
ANI(1,2)=-AGM/AA
ANI(2,1)=-AGM/AA
ANI(2,2)=1./AA
ANI(3,2)=ANI(2,2)
ANI(3,3)=ANI(2,1)
ANI(4,2)=-ANI(2,1)
ANI(4,3)=ANI(1,1)
ANI(5,1)=ANI(1,1)
ANI(5,3)=ANI(1,2)
ANI(6,1)=ANI(1,2)
ANI(6,3)=ANI(1,1)

DO 100 J=1,NYP
000 ENJ(J)=BB/AA

GO TO 800

800 CONTINUE
GO TO (1,2),I(1)

***** ANDEKREGEN MODEL FOR LINEARIZATION MODEL *****
1-CONTINUE
S=DSIN(PII(1))
C=DCOS(PII(1))
CC=1./C
ANI(1,1)=1.-S
ANI(1,2)=-S
ANI(1,3)=-1.-S
ANI(1,4)=-1.-S
ANI(1,5)=-S
ANI(1,6)=1.-S
ANI(2,1)=-1.-S
ANI(2,2)=-S
ANI(2,3)=1.-S
ANI(2,4)=1.-S
ANI(2,5)=-S
ANI(2,6)=-1.-S
ANI(3,1)=2/1.73
ANI(3,2)=1/1.73
ANI(3,3)=ANI(3,1)
ANI(3,4)=-ANI(3,1)
ANI(3,5)=-ANI(3,2)
ANI(3,6)=-ANI(3,1)

```

WRITE(6,750)
750 FORMAT(1H1Z' **** PRIN OF MATRICES OF DIMEN *** ' /)
WRITE(6,751) ((ANG(I,J),J=1,NXP),I=1,NXC)
751 FORMAT(5X,17.3)
DO 1200 J=1,NXP
DO 1200 I=1,NXC
1200 ANG(J,I)=ANG(I,J)+CC
DO 1000 J=1,NXP
DO 1000 I=1,NXC
1000 ANI(J,I)=ANG(J,I)

WRITE(6,752) ((ANG(J,I),I=1,NXC),J=1,NXP)
WRITE(6,752) ((ANI(J,I),I=1,NXC),J=1,NXP)
752 FORMAT(5X,3F17.3)
DO 1100 J=1,NXP
1100 FRJ(J)=Z.*(COS(B)

GO TO 1500

2 CONTINUE
X=(Z.*PI)/NXP
Y=PI/NXP

DO 3 KP=1,NXP
ANI(KP,1)=DCOS(X*KP)+(DSIN(FBI(N))*DCOS(Y))
ANI(KP,2)=FCOS(X*KP)+(DSIN(FBI(L))*DCOS(Y))

ANI(KP,2)=-ANI(KP,2)

ANI(KP,3)=Z.*(DSIN(X*KP))
FRJ(KP)=Z.*(COS(B)*(DCOS(FBI(L)))+(DCOS(Y))

3 CONTINUE

900 CONTINUE

IF (IPD.AX.EQ.0.OR.IPBL.AX.EQ.1) GO TO 1500

ANI(1,1)=1.0
ANI(1,3)=1.0
ANI(2,1)=1.0
ANI(2,3)=-1.0
ANI(3,1)=-1.0
ANI(3,3)=1.0
ANI(4,1)=-1.0
ANI(4,3)=-1.0
ANI(5,2)=1.0
ANI(5,3)=1.0
ANI(6,2)=1.0
ANI(6,3)=-1.0
ANI(7,2)=-1.0
ANI(7,3)=1.0
ANI(8,2)=-1.0
ANI(8,3)=-1.0
ANI(9,1)=1.0
ANI(9,3)=1.0

```

```

ANI(9,2)=-1.0      00177**8
ANI(9,3)=1.0      00178**8
ANI(10,1)=1.0     00179**8
ANI(10,2)=-1.0   00180**8
ANI(10,3)=-1.0   00181**8
ANI(11,1)=-1.0   00182**8
ANI(11,2)=1.0    00183**8
ANI(11,3)=1.0    00184**8
  ANI(12,1)=-1.0  00185**8
ANI(12,2)=1.0    00186**8
ANI(12,3)=-1.0   00187**8
ANI(13,3)=1.0    00188**8
ANI(14,3)=-1.0   00189**8
ND=NYP-2         00190**9
WRITE(6,1700) IPLNAX,COH(N) 00191*10
1700 FORMAT(5X,15,E10.3) 00192*10
                                00193**9
                                00194**9
                                00195**9
  DO 1800 J=1,ND 00196**8
1800 EKJ(J)=COH(N) 00197**9
      EKJ(NYP-1)=COH(N)/1.73 00198*19
      EKJ(NYP)=EKJ(NYP-1) 00199**9

      GO TO 1500

```

```

655 CONTINUE
***** AXISYMMETRIC PROBLEM *****

```

```

1500 CONTINUE 00200**8
      WRITE(10) ((ANI(J,I),I=1,NSC),J=1,NYP),(EKJ(J),J=1,NYP) 00201**8
                                00202**2
                                00203**2
                                00204**2
      WRITE(6,502) 00205
      DO 504 J=1,NYP 00206
504 WRITE(6,503)J,(ANI(J,I),I=1,NSC),EKJ(J) 00207
                                00208**4
                                00209**4
501 FORMAT(4E10.3) 00210
502 FORMAT(/43HINPUT TABLE 2B. LINEARIZED YIELD FUNCTION //5X, 00211
      16H YIELD,5X,27HDIRECTION COSIN OF THE UNIT,5X,13HDISTANCE FROM / 00212
      25X,6H PLANE,19X,13HNORMAL VECTOR,6X,12HSTRESS ORIGIN/) 00213
503 FORMAT(5X,16,2X,3F10.3,8X,F10.3) 00214
300 CONTINUE 00215**2
                                00216**2
                                00217**2
                                00218**2
      RETURN 00219
      END 00220

```

```

DATA SET YLDRA      AT LEVEL 020 AS OF 83/02/11
SUBROUTINE YLDRA(NEL,NYP,NSC,ANI,AMJ,LC,NLC,AM,EK,AN,NES,AMAX,
*ERD,SIG,IR,MP1,IP1,VOL)
IMPLICIT REAL*8 (A-H,O-Z)

COMMON/BOBBY/KKS,NUMGT,NC,LOOP,NDRH,NINI,LDWA,NFT
COMMON/CONTR/ID,HED(8),RADN,NASLM,NUMSL,NKK,MBAND,NCYCL
*,NDYN,IP1,MAX,ISIRPK,NKRINT,ISTART,NEGH,MBANDH,NEGS,MBANDS
COMMON/BERNYS/SG(5,200,4,4),NCOND
COMMON/DIMEL/N43,KP1,KP2,KK,NPRNT

DIMENSION ANI(NYP,1),AMJ(14),AM(1),EK(1),AN(NES,1)
DIMENSION AMJ(3),AMAX(NYP,1,1)
DIMENSION ERD(1),SIG(9)
00001**3
00002*11
00003*19
00004*10
00005
00006*10
00007*10
00008*18
00009*18
00010*16
00011*20
00012*20

00013*10
00014*16
00015*18
00016**4
00017
*****      N * TRANSPOSED * SIGMA CALCULATION *****00018
COMPUTE THE MAXIMUM WITH RESPECT TO THE TIME OF THE STRESS COMPONANT00019
NORMAL TO EACH YIELD PLANE FOR EACH ELEMENT.      00020
00021
NDT=1
IF(NCOND.EQ.1) NDT=4
M=NEL
DO14 J=1,NYP
AMJ(J)=0.0
DO11 I=1,NSC
ANI(I)=ANI(J,I)
IF(1.EQ.NSC) GO TO 5
AMJ(J)=AMJ(J)+ANI(I)*SIG(I)
GO TO 4
5 AMJ(J)=AMJ(J)+ANI(I)*SIG(I+1)
4 CONTINUE
00031
00032**8
*****      IMPOSE ARTIFICIAL CONDITION SO THAT STRESS00033*12
ARE ALWAYS IN COMPRESSTION *****00034*12
00035*12
00036*12
00037*12
00038**8
*****      MODIFY FOR VERTICAL LOADING LOAD CASE NO. 300039**8
00040**8
00041*17
IF(LC.EQ.NLC) GO TO 14
00042*17
00043*12
*****      SET ABSOLUTE VALUES FOR MAXIMA CHECKING *****00044*12
00045*12
AND=AMJ(J)
00046*14
ANG=AMAX(J,M,IP1)
00047*18
00048*18
00049*18
IF(AND.LT.ANG) GO TO 14
00050*12
00051*18
AMAX(J,M,IP1)=AMJ(J)
00052*18
00053*18

```

```

14 CONTINUE                                00054
                                           00055
                                           00056
ASSEMBLE THE MAXIMUM STRESS COMPONENT IN ONE VECTOR AM ,THE NORMAL 00057
AND THE NORMALITY MATRIX AN.                                00058
DISTANCE FROM STRESS ORIGIN TO THE YIELD PLANES IN ONE VECTOR EK , 00059
                                           00060
IF (DC.LI.NC.AND.NC.LI.NCYCL) GO TO 11 00061*10
DO 6 J=1,NYP                                     00062
LX=J+(M-1)*NYP*NDI+(IPT-1)*NYP                00063*18
AM(LX)=AMAX(J,M,IPT)                            00064*18
                                           00065*18
                                           00066*18
                                           00067**8

IF (KPT.EQ.1) GO TO 15

*****      MODIFY K- VECTOR FOR VERTICAL LOADING      ***** 00068**8
EK(LX)=(EKJ(J)-AMJ(J))                          00069**8
GO TO 6                                           00070**8
                                           00071*15

15 CONTINUE
EK(LX)=EKJ(J)

6 CONTINUE                                00072**5
                                           00073**8
                                           00074**8
DO 200 J=1,NYP                                   00075**8
LX=J+(M-1)*NYP*NDI+(IPT-1)*NYP                00076**9
DO 100 I=1,NSC                                  00077*18
LY=1+(M-1)*NSC*NDI+(IPT-1)*NSC                00078*18
100 AN(LY,LX)=ANI(J,I)                          00079**9
200 CONTINUE                                    00080*18
11 CONTINUE                                    00081**9
                                           00082**9
                                           00083
RETURN                                          00084
END                                             00085
                                           00086

```


SUBROUTINE ZAMIN(CB,A,AM,C,AM,EN,GB,SU,SU,ICOLMS,IDES,ROW,CUPI,
*PSOL,DSOL,G,IBR,NPK,NPJ,NEQS,NUMEL,NYP,NSC,SU,SN,CT,IH,NUMNP)

***** DIMENSION *****

IMPLICIT REAL*8(A-H,O-Z)

COMMON/TARA/LPC,N,M1,M2,NPT,KPT,NPKNT,N,MM

DIMENSION CE(1),A(MM,1),AN(NPJ,1),C(NPJ,1),AM(1),EK(1),GB(1),SU(1)
*,SU(1),ICOLMS(1),IDES(1),ROW(1),CUPI(MM,1),PSOL(1),DSOL(1),G(1)

DIMENSION SD(NPJ,1),SN(1),CT(NPJ,1),IH(NUMNP,1)

95 CONTINUE

WRITE(6,96) LPC,M1,M2,N,NPT,KPT,NPKNT,N,MM,NPJ,NPK,NEQS

96 FORMAT(16I5)

IF(NPT.EQ.2) GO TO 14

10 DO 10 J=1,NPK
A(LPC,J)=AM(J)

NO=NPK+1
DO 11 J=NO,N
11 A(LPC,J)=0.00

33 DO 33 I=1,NEQS
SU(1)=0.0

GO TO 18

14 CONTINUE
JJ=NPK+1
JJ1=NPK+NEQS

19 DO 19 J=1,NPK
A(LPC,J)=0.00
DO 20 J=JJ,JJ1
20 A(LPC,J)=GB(J-NPK)
JJ2=NPK+NEQS+1
DO 21 J=JJ2,N
21 A(LPC,J)=-GB(J-JJ1)

18 CONTINUE

DO 1 I=1,NPK
A(2,I)=+EK(I)
1 CONTINUE

```

C      WRITE(6,200) (A(2,1),I=1,NPK)
      N1=NPK+1
      N2=NPK+NEQS
e      DO 2 I=N1,N2
2      A(2,1)=-S0(1-NPK)
      N3=NPK+NEQS+1
      N4=N
      DO 3 I=N3,N
3      A(2,1)=+S0(1-N2)
C
C      DO 4 I=1,N
      SUM=0.0
      DO 5 J=3,LPC
5      SUM=SUM+A(0,I)
      G(I)=-A(2,1)+SUM
4      CONTINUE
34      WRITE(6,34) (S0(I),I=1,NEQS)
      FORMAT(1ZE10.3)
C
      I1=1
C
      DO 6 I=1,N
6      A(I1,I)=G(I)
C
      DO 8 I=1,LPC
8      S0(I)=0.0
      S0(LPC)=1.00
C
      ICOLMS(I1)=2
      ICOLMS(I1+1)=1
      DO 9 I=3,LPC
9      ICOLMS(I)=1
C
      ROW(I1)=-1.00
      ROW(I1+1)=1.00
C
      DO 13 I=3,LPC
13      ROW(I)=0.00
C
      KG=1
      ITHAX=5*N
      IR=M+1
C
      DO 56 I=1,MM
      DO 12 J=1,MM
      COPI(I,J)=0.0
12      CONTINUE
56      CONTINUE
      DO 17 I=1,MM
      COPI(I,1)=1.00
17      CONTINUE

```

```
C
C      COPI(I,I+1)=1.00
C
```

```
      IDES(I)=N+2
      IDES(I+1)=N+1
      DO 140 I=3,LPC
140 IDES(I)=N+1
```

```
C
C      WRITE(6,100)
100 FORMAT(1H17,1,PRINT,*)
      WRITE(6,200) ((A(I,J),J=1,12),I=1,15)
      WRITE(6,999)
      WRITE(6,200) ((A(I,J),J=13,24),I=1,15)
      WRITE(6,999)
      WRITE(6,210) ((A(I,J),J=25,32),I=1,15)
      WRITE(6,999)
      WRITE(6,210) ((A(I,J),J=33,40),I=1,15)
C
```

```
200 FORMAT(5X,12E10.3)
210 FORMAT(5X,8E10.3)
      WRITE(6,100)
```

```
      WRITE(6,200) (SQ(I),I=1,LPC)
      WRITE(6,999)
```

```
999 FORMAT(//)
      WRITE(6,150) (ICULMS(I),I=1,LPC)
150 FORMAT(10I5)
```

```
      WRITE(6,999)
      WRITE(6,200) (ROW(I),I=1,LPC)
      WRITE(6,999)
      WRITE(6,200) ((COPI(I,J),J=1,12),I=1,MM)
      WRITE(6,999)
      WRITE(6,150) (IDES(I),I=1,LPC)
```

```
      IPHASE=1
```

```
      CALL ZAOLP(IPHASE,A,SQ,ICULMS,ROW,KG,M,N,ITMAX,LPC,MM,COPI,IDES,
+PSUB,DSUB,IER)
```

```
      WRITE(6,91) IER,ITMAX
91 FORMAT(5X,2I5)
```

```
      IF(IER.NE.0) GO TO 16
```

```
      DO 15 I=2,LPC
```

```
15 IF(IDES(I).GT.(N+1)) GO TO 29
```

```
      WRITE(6,999)
      WRITE(6,150) (IDES(I),I=1,LPC)
```

```

C
C
C
C      WRITE(6,999)
C      WRITE(6,150) (ICOLMS(I),I=1,LPC)
C
C      LPC=1
C      DO 22 I=1,LPC
22  ICOLMS(I)=ICOLMS(I)+ISIGN(I,ICOLMS(I))
C
C      ROW(11)=1.00
C      M=M-1
C
C      IPHASE=Z
C      I=1
C      I=Z
C
C      CALL ZAOI(IPHASE,A(I,1),SQ(I),ICOLMS,ROW,KG,M,N,ITMAX,LPC,MM,
C      *COPI(I,1),IDES(I),PSOL,DSOL,IER)
C
C      WRITE(6,91) IER,ITMAX
C
C      DO 23 I=1,N
23  G(I)=0.0
C
C      N=NZ+1
C      DO 24 I=1,N
24  IF (IDES(I+1).EQ.0) Z=-PSOL(I)
      IF (IDES(I+1).NE.0) G(IDES(I+1))=PSOL(I)
      CONTINUE
C
C      DO 25 I=1,NEQS
25  G(1+NEK)=G(1+NEK)-G(1+NEK+NEQS)
      CONTINUE
C
C      CE(1)=Z
C
C      DO 26 I=1,NEK
26  CE(1+I)=G(1)
C
C      DO 27 I=1,NEQS
27  CE(1+NEK+I)=G(1+NEK)
C
C      NG=NEK+NEQS+1
C      WRITE(6,52)
C      FORMAT(1H7'-PRINT OF ZAOI')
C      WRITE(6,53) (CE(I),I=1,NG)
C      FORMAT(5X,12E10.3)
C
C
C      CALL ODFDI(GOINEL,ETP,ESC,DEL,NEK,CE,Z,AN,SD,SN,CF,SO,NEQS,IR,
C      *RUMRP)
C      GO TO 55
C
29  CONTINUE
C
      WRITE(6,50)

```

C 50 FORMAT(IH1)' PHASE1 DID NOT REMOVE ALL ARTIFICIAL VARIABLES '

C 10 CONTINUE

C 49 WRITE(0,49) PHASE ERROR SET'

C 55 CONTINUE

RETURN
END

