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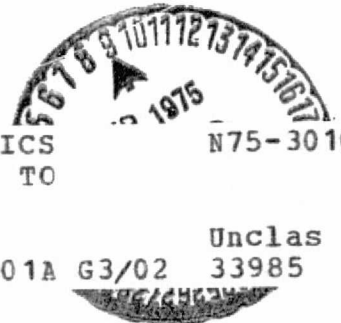
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THREE-DIMENSIONAL DYNAMICS OF
SCIENTIFIC BALLOON SYSTEMS IN RESPONSE TO
SUDDEN GUST LOADINGS

by Daniel R. Dorsey, Jr.

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THREE-DIMENSIONAL DYNAMICS OF SCIENTIFIC
BALLOON SYSTEMS IN RESPONSE TO SUDDEN GUST LOADINGS

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SUMMARY

A mathematical model of the three-dimensional dynamics of a high-altitude scientific research balloon system perturbed from its equilibrium configuration by an arbitrary gust loading is developed. The platform is modelled as a system of four coupled pendula. The equations of motion are developed in the Lagrangian formalism assuming a small-angle approximation. Three-dimensional pendulation, torsion and precessional motion due to Coriolis forces are considered. Aerodynamic and viscous damping effects on the pendulatory and torsional motions are included.

A general model of the gust field incident upon the balloon system is also developed.

The report is concluded with a description of the digital computer simulation program and a guide to its use.

INTRODUCTION

The use of high-altitude balloon systems as stable platforms for scientific studies of the earth's atmosphere requires a knowledge of the dynamics of the system for proper data interpretation. This knowledge is especially critical in studies of the vertical distributions of atmospheric trace constituents and temperature, since data interpretation is highly sensitive to the attitude of the data-collecting instrument. It is well known that most systems remain stable at float altitude, but we need a mathematical model which can predict the attitude and motion of the system with sufficient accuracy to assure accurate interpretation and reduction of data.

In reality, the motion of a perturbed balloon system at float

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altitude is extremely complex, involving interactions between several different types of oscillation such as bounce, pendulation, rotation and precession. Further complexity may be introduced by such factors as damping (boundary layer) forces, air turbulence, cable elasticity and motion induced by the support mechanisms of the system (i.e., independent motion of a particular system element).

A number of models have been proposed to account for some of these factors. Morris and Stefan (ref. 1) have studied plane pendulation, bounce, plane torsional motion (including damping) and the atmospheric environment for a two element system (balloon and gondola). Emslie, et.al. (ref.2) have analytically and experimentally considered balloon bounce, the thermodynamics of the enclosed gas, and heat transfer between atmosphere, balloon fabric and enclosed gas. The azimuthal rotations of a balloon-gondola system have been studied in detail, both theoretically and experimentally, by Germeles, et.al. (ref.3), with particular emphasis on the aerodynamic damping and inertia of the system. Further study of the aerodynamic parameters of a rotating sphere has been done by several investigators (refs.4,5,6,7,8 and 9).

To date, there appears to have been no serious effort to develop a detailed general model defining the gross motions of a balloon system in three dimensions. This report is intended to develop such a model. Admittedly, the model will not be completely general, but as further work is done on this subject, we may expect further sophistication and generality in a three-dimensional model. The balloon system considered in this report is the NASA-Langley/AFGRL LACATE (Lower Atmosphere Composition And Temperature Experiment) system which is to be flown at 150 km altitude. The system consists of a 39 million cubic foot capacity balloon, a "load bar" containing balloon control equipment and cable reel, a gondola recovery parachute package and a research payload gondola. This system is configured as a system of four coupled pendula.

A cursory derivation of the equations of motion (EOM) for plane damped pendulation for this system was done by Hinton of NASA-Langley. This report will investigate (1) three-dimensional pendulation, (2) torsional motion, (3) aerodynamic and viscous damping effects on pendulatory and torsional motion, and (4) precessional motion due to Coriolis forces. We shall also attempt

to develop a detailed general model for gust loading in three dimensions and study possible gust loading configurations. Since we are concerned primarily with the gross motions of the system, we will not investigate such factors as cable elasticity, statistical variation or turbulence in the gust field, bounce or atmospheric thermodynamics. This is done in the interest of keeping the analysis mathematically tractable.

SYMBOLS

a	balloon-to loadbar cable length = 72 feet
A_k	surface area of kth element
$[A]$	dynamic coupling coefficient matrix
$[B]$	static coupling coefficient matrix
c	parachute package-to-payload cable length = 5 feet
C_k	drag coefficient of kth element
$[D]$	damping matrix
f	loadbar-to-parachute package cable length = 1310 feet
F_{cor}	Coriolis force
g	gravitational acceleration at float altitude
h_k	length of kth cylinder
I_k	moment of inertia of kth element
J	impulsive torque
m_k	mass of kth element
m_1	m_e
m_2	$m_e + m_p$
m_3	$m_e + m_p + m_s$
m_4	$m_e + m_p + m_s + m_b$
N	torque
P_r	linear impulse
q_k	generalized coordinate of kth element
Q_k	nonconservative force on kth element
r_p	balloon radius = 210 feet

R_k	radius of kth cylinder
t	time
T	kinetic energy
v_k	linear velocity of kth element
V	potential energy
x,y,z	Cartesian coordinates
ζ	distance from element center of mass (azimuth plane)
η_k	damping constant of kth element
θ_k	deflection angle off vertical, kth element
λ	geocentric latitude
μ	viscosity of air at float altitude
ρ	density of air at float altitude
τ_k	torsional constant of kth element
$[\tau]$	torsional coupling coefficient matrix
ϕ_k	azimuthal deflection angle, kth element
χ	angle of rotation about azimuth plane
ψ	gust axis incidence angle relative to horizontal
ω_0	rotation rate of earth
ω_k	angular velocity of kth element produced by gust

SUBSCRIPTS

1	initial value
2	final value
k	system element (b,s,p or e)
b	balloon
s	load bar
p	parachute package
e	payload package

FORMULATION OF PROBLEM

The analysis will be carried out using a spherical polar coordinate system and an idealized model of the balloon system as shown in Figure 1. The origin of this coordinate system is taken to coincide with the geometric center of the balloon. The

z-axis is taken as the local vertical and is positive downward. Similar systems are set up individually for the load bar, parachute package and gondola, each with the z-axis as the local vertical and the same positive sense as the balloon z-axis. This system facilitates the definition of angular deflections for each element of the system, since, proceeding down from the balloon, the total angular deflection of the system down to the element in question is additive with respect to the balloon center. This also makes algebraic work considerably more manageable.

With regard to the system model, we make some basic idealizations. Since the moment of inertia of the balloon alone is much greater than that of any of the other system elements, the balloon is assumed to be a spherical shell with an extremely small thickness-to-radius ratio (ref.3). Elastic deformation of this shape due to gust loading will be neglected. Secondly, we idealize the load bar, parachute package and gondola as right circular cylinders; these are not necessarily their actual shapes, but they will be defined as cylinders with moments of inertia equivalent to the actual shape to simplify rigid body computations. Analysis in a Eulerian system was considered, but was deemed too difficult to attempt for the problem at hand; the Euler approach may, however, be more general and should be studied in the future.

The equations of motion are developed in the Lagrangian formalism. Since angular deflections due to gust loading are anticipated to be no greater than 0.01 radians during flight (ref.10), the small-angle approximation is invoked to simplify the mathematics somewhat. We will not consider bounce in this report; translational motion of the balloon center of mass (CM) is included while assuming the balloon CM to remain at its initial equilibrium (float) altitude. Thus, we must deal with a system having eight degrees of freedom. The θ and ψ deflections and their corresponding angular velocities are found for each system element individually; obviously, moving down from the balloon toward the payload, the gross motion of the system becomes more complex due to the motions of the elements above the one under study. Pendulation (with and without damping), torsional (azimuthal) motion (with and without damping) and precessional motion will be considered separately. In this manner, we will be able to examine

the character of each type of motion alone; then, when we consider various combinations of motion, we will have some means of judging the degree to which each type of motion contributes to the overall motion of the system.

The Lagrangian EOM are derived in the standard form

$$\frac{d}{dt} \frac{\partial T}{\partial \dot{q}_k} - \frac{\partial T}{\partial q_k} + \frac{\partial V}{\partial q_k} = Q_k \quad (1)$$

where, for the problem at hand, the generalized coordinates q_k are the eight angles $\theta_b, \varphi_b, \theta_s, \varphi_s, \theta_p, \varphi_p, \theta_e$ and φ_e . The generalized nonconservative forces Q_k take into account the aerodynamic and viscous damping terms. We should note at this point that the EOM developed in the following sections are not carried to a closed form analytical solution. Solutions could be found in closed form in principle, but the algebraic work required to do so would be prohibitive. Hence, the EOM are developed to the point where they are easily integrable numerically. Terms due to separate motion types may be added together and then integrated as a whole to obtain the overall motion.

After the EOM are derived, we then address the problem of defining the gust loading model, since the gust velocity field configuration incident on the system will define the initial conditions for solution of the EOM. No attempt will be made in this report to define the stability criteria for system motions; this problem will be taken up in a future study.

DEVELOPMENT OF THE EQUATIONS OF MOTION

Pendulatory Motion

We first consider pendulation without the inclusion of damping terms. The total kinetic and potential energies are

$$T = \frac{1}{2} \sum_k [m_k v_k^2 + I_k (\dot{\theta}_k^2 + \dot{\varphi}_k^2)] \quad , \quad k = b, s, p \text{ and } e \quad (2)$$

and

$$V = m_s g r_b (1 - \cos \theta_b) + a(1 - \cos \theta_s) + m_p g r_b (1 - \cos \theta_b) + a(1 - \cos \theta_s) + f(1 - \cos \theta_p) + m_e g r_b (1 - \cos \theta_b) + a(1 - \cos \theta_s) + f(1 - \cos \theta_p) + c(1 - \cos \theta_e) \quad (3)$$

The linear velocities v_k are first defined in a Cartesian coordinate system by their x,y,z components and then transformed to the spherical polar representation. In Cartesian coordinates

$$v_{b_i} = \alpha_i \quad , \quad v_b = v_{b_x} + v_{b_y} + v_{b_z} \quad (4)$$

$$v_{s_i} = \alpha_i + \beta_i \quad , \quad v_s = v_{s_x} + v_{s_y} + v_{s_z} \quad (5)$$

$$v_{p_i} = \alpha_i + \beta_i + \gamma_i \quad , \quad v_p = v_{p_x} + v_{p_y} + v_{p_z} \quad (6)$$

$$v_{e_i} = \alpha_i + \beta_i + \gamma_i + \delta_i \quad , \quad v_e = v_{e_x} + v_{e_y} + v_{e_z} \quad (7)$$

where $i = x, y, z$ successively. Transformation to spherical polar coordinates is made through the variables $\alpha, \beta, \gamma, \delta$. For the balloon we have

$$\begin{aligned} \alpha_x &= r_b (\dot{\theta}_b \cos \theta_b \cos \varphi_b - \dot{\varphi}_b \sin \theta_b \sin \varphi_b) \\ \alpha_y &= r_b (\dot{\theta}_b \cos \theta_b \sin \varphi_b + \dot{\varphi}_b \sin \theta_b \cos \varphi_b) \\ \alpha_z &= -r_b \dot{\theta}_b \sin \theta_b \end{aligned} \quad (8)$$

The transformation relations for the load bar (β), parachute package (γ) and gondola (δ) are of the same form as Equation (8), with r_b replaced by a, f, c and the angle subscripts replaced by s, p, e respectively.

With T and V now completely specified, the EOM can be developed from Equation (1). Taking the appropriate derivatives and after considerable algebraic manipulation, we obtain eight EOM corresponding to the eight degrees of freedom $\theta_b, \varphi_b, \theta_s, \varphi_s, \theta_p, \varphi_p, \theta_e$ and φ_e . We find that these EOM are represented in the format of a system of eight simultaneous linear equations, i.e.,

$$\begin{bmatrix} A_{11} & A_{12} & \cdot & \cdot & \cdot & A_{18} \\ A_{21} & A_{22} & \cdot & \cdot & \cdot & A_{28} \\ A_{31} & \cdot & \cdot & \cdot & \cdot & \cdot \\ A_{41} & \cdot & \cdot & \cdot & \cdot & \cdot \\ A_{51} & \cdot & \cdot & \cdot & \cdot & \cdot \\ A_{61} & \cdot & \cdot & \cdot & \cdot & \cdot \\ A_{71} & \cdot & \cdot & \cdot & \cdot & \cdot \\ A_{81} & A_{82} & \cdot & \cdot & \cdot & A_{88} \end{bmatrix} \begin{bmatrix} \ddot{\theta}_b \\ \ddot{\varphi}_b \\ \ddot{\theta}_s \\ \ddot{\varphi}_s \\ \ddot{\theta}_p \\ \ddot{\varphi}_p \\ \ddot{\theta}_e \\ \ddot{\varphi}_e \end{bmatrix} = \begin{bmatrix} B_1 \\ B_2 \\ \cdot \\ \cdot \\ \cdot \\ \cdot \\ \cdot \\ B_8 \end{bmatrix} \quad (9)$$

or in more compact matrix notation

$$[A][\ddot{q}] = [B] \quad (10)$$

where: $[A]$ is the dynamic coupling coefficient matrix containing the coefficients of all terms involving the second derivative with respect to time, and $[B]$ is the static coupling coefficient matrix containing all other terms (including damping terms). The elements of $[A]$ and $[B]$ are given in the Appendix A.

It is clear that the pendulation EOM exhibit rather strong dynamic coupling for the entire balloon system; $[B]$ indicates the presence of static coupling as well. This is not surprising, considering the complexity of the problem. Equation (10) may be solved for the desired solution matrix $[\ddot{q}]$ using the Gaussian elimination algorithm (or method of detached coefficients¹¹), and the resulting equations integrated directly to obtain $[\dot{q}]$ and $[q]$. The extremely high degree of dynamic and static coupling, however, makes direct analytic closed-form solution and integration prohibitively cumbersome as far as algebraic work is concerned.

Torsional Motion

The torsional EOM derivation will be confined to motion in the azimuth plane. This is done to avoid using a Eulerian angle formalism, for the same reasons as mentioned before. This is not restrictive for our purposes; since we use a small-angle approximation, motion in the θ -direction will be small compared to the motion in the azimuth plane. When pendulation and torsion are considered together, the torsional motion may be considered superimposable onto the pendulatory motion.

The kinetic and potential energies are written^{12,13}

$$T = \frac{1}{2}(I_b \dot{\phi}_b^2 + I_s \dot{\phi}_s^2 + I_p \dot{\phi}_p^2 + I_e \dot{\phi}_e^2) \quad (11)$$

$$W = \frac{1}{2} \tau_s (\phi_b - \phi_s)^2 + \tau_p (\phi_s - \phi_p)^2 + \tau_e (\phi_p - \phi_e)^2$$

where: the torsion constants τ_k are defined by¹

$$\tau_s = -m_s g R_s^2 / (r_b + a) \quad ; \quad \tau_p = -m_p g R_p^2 / (r_b + a + f + h_s) \quad (12)$$

$$\tau_e = -m_e g R_e^2 / (r_b + a + f + c + h_s + h_p)$$

The Lagrangian EOM for the undamped case are¹²

$$\begin{aligned}
 I_b \ddot{\varphi}_b + \tau_s (\varphi_b - \varphi_s) &= 0 \\
 I_s \ddot{\varphi}_s - \tau_s (\varphi_b - \varphi_s) + \tau_p (\varphi_s - \varphi_p) &= 0 \\
 I_p \ddot{\varphi}_p - \tau_p (\varphi_s - \varphi_p) + \tau_e (\varphi_p - \varphi_e) &= 0 \\
 I_e \ddot{\varphi}_e - \tau_e (\varphi_p - \varphi_e) &= 0
 \end{aligned}
 \tag{13}$$

In order to include viscous and aerodynamic damping effects, we define the damping constants η_k ¹

$$\begin{aligned}
 \eta_b &= 2.5(\rho\mu)^{\frac{1}{2}} R_b^4 |\dot{\varphi}_b|^{\frac{1}{2}} & \eta_s &= 2.5(\rho\mu)^{\frac{1}{2}} R_s^4 |\dot{\varphi}_s|^{\frac{1}{2}} \\
 \eta_p &= 2.5(\rho\mu)^{\frac{1}{2}} R_p^4 |\dot{\varphi}_p|^{\frac{1}{2}} & \eta_e &= 2.5(\rho\mu)^{\frac{1}{2}} R_e^4 |\dot{\varphi}_e|^{\frac{1}{2}}
 \end{aligned}
 \tag{14}$$

so that the EOM for the damped case are

$$\begin{aligned}
 I_b \ddot{\varphi}_b &= -\eta_b \dot{\varphi}_b - \tau_s (\varphi_b - \varphi_s) \\
 I_s \ddot{\varphi}_s &= -\eta_s \dot{\varphi}_s + \tau_s \varphi_b - (\tau_p + \tau_s) \varphi_s + \tau_p \varphi_p \\
 I_p \ddot{\varphi}_p &= -\eta_p \dot{\varphi}_p + \tau_p \varphi_s - (\tau_p + \tau_e) \varphi_p + \tau_e \varphi_e \\
 I_e \ddot{\varphi}_e &= -\eta_e \dot{\varphi}_e + \tau_e (\varphi_p - \varphi_e)
 \end{aligned}
 \tag{15}$$

The EOM for both the undamped and damped cases show the existence of only static coupling; hence the system of equations in both cases may be integrated directly. Note also that for each case a system of only four EOM is obtained, in keeping with our consideration of motion in the azimuth plane.

In the digital computer program, the EOM are solved in the forms given by Equations (13) and (15). Analytic solutions to these EOM may be obtained by assuming solutions of the form $\varphi_i = R_i e^{kt}$, $i=b,s,p,e$. For nontrivial solutions, the EOM must satisfy the condition

$$\sum_{i=1}^8 R_i k_i = 0 \tag{16}$$

where the f_i are constant coefficients of the k^i ; this equation is of eighth degree in k and thus has eight roots. Since the $f_i > 0$, there exist no real roots > 0 , but real roots < 0 may exist. Complex roots are also admissible and would occur in conjugate pairs. Since we require $\varphi_i \rightarrow 0$ as $t \rightarrow \infty$, the real parts of complex roots must be < 0 . If we assume the existence of α complex roots of the form $k=p+iq$ and β real negative roots $u(u > 0)$, we may write the solutions as

$$\varphi_i = \sum_{j=1}^{\alpha} F_{ij} e^{-p_j t} \sin(q_j t + \phi_{ij}) + \sum_{m=1}^{\beta} G_{im} e^{-u_m t} \quad (17)$$

so that solutions can be constructed once the k_i are known, but the algebraic work would become extremely cumbersome. That sinusoidal solutions of this form do indeed exist is verified by the computer program results, which will be discussed later.

In order to superimpose torsional motion upon the pendulatory motion, we must define the analogous forms of the matrices $[A]$ and $[B]$ as defined in Equation (9). We set $A_{i1}=A_{i3}=A_{i5}=A_{i7}=0$ and $B_1=B_3=B_5=B_7=0$; also, $A_{i2}=I_p, A_{i4}=I_s, A_{i6}=I_p, A_{i8}=I_e$. B_2, B_4, B_6 and B_8 will correspond to the right hand sides of the EOM; thus, the elements B_i corresponding to torsional terms can be added to the pendulation $[B]$ by introducing a new eight-element torsional coupling coefficient matrix $[\tau]$, which is also a column matrix.

Inclusion of Damping Effects

Damping forces are nonconservative in nature and are included in the Q_k term of Equation (1). The work of this section is applicable to both pendulatory and torsional motion. Following the lead of Germeles, et al.³, we assume viscous damping on the balloon proper, and with spherical symmetry for the balloon, we have

$$\begin{aligned} Q_{b\theta} &= 2.5(\rho\mu)^{\frac{1}{2}} r_b^3 |\dot{\theta}_b|^{\frac{1}{2}} \dot{\theta}_b \\ Q_{b\dot{\theta}} &= 2.5(\rho\mu)^{\frac{1}{2}} r_b^3 |\dot{\varphi}_b|^{\frac{1}{2}} \dot{\varphi}_b \end{aligned} \quad (18)$$

Aerodynamic damping is assumed to affect the other system elements, and is derived from the definition of the aerodynamic drag coefficient. Drag on supporting cables is neglected, since

the surface area of the cables is small compared to that of the other system elements. Hence

$$\begin{aligned}
 Q_s &= \frac{1}{2} \rho A_s C_s v_s^2 (r_b + a) &= \frac{1}{2} A_s C_s v_s v_s (r_b + a) \\
 Q_p &= \frac{1}{2} \rho A_p C_p v_p^2 (r_b + a + f) &= \frac{1}{2} A_p C_p v_p v_p (r_b + a + f) \\
 Q_e &= \frac{1}{2} \rho A_e C_e v_e^2 (r_b + a + f + c) &= \frac{1}{2} A_e C_e v_e v_e (r_b + a + f + c)
 \end{aligned} \tag{19}$$

To be properly added to the matrix $[B]$, Equations (19) must be separated into their θ and φ components. Transforming the v_k by the relations¹⁴

$$\begin{aligned}
 v_{s\theta} &= r_b \dot{\theta}_b + a \dot{\theta}_s & v_{s\varphi} &= r_b \dot{\varphi}_b + a \dot{\varphi}_s \\
 v_{p\theta} &= r_b \dot{\theta}_b + a \dot{\theta}_s + f \dot{\theta}_p & v_{p\varphi} &= r_b \dot{\varphi}_b + a \dot{\varphi}_s + f \dot{\varphi}_p \\
 v_{e\theta} &= r_b \dot{\theta}_b + a \dot{\theta}_s + f \dot{\theta}_p + c \dot{\theta}_e & v_{e\varphi} &= r_b \dot{\varphi}_b + a \dot{\varphi}_s + f \dot{\varphi}_p + c \dot{\varphi}_e
 \end{aligned} \tag{20}$$

we then have

$$\begin{aligned}
 Q_{s\theta} &= \frac{1}{2} \rho A_s C_s (r_b + a) (r_b \dot{\theta}_b + a \dot{\theta}_s)^2 \\
 Q_{p\theta} &= \frac{1}{2} \rho A_p C_p (r_b + a + f) (r_b \dot{\theta}_b + a \dot{\theta}_s + f \dot{\theta}_p)^2 \\
 Q_{e\theta} &= \frac{1}{2} \rho A_e C_e (r_b + a + f + c) (r_b \dot{\theta}_b + a \dot{\theta}_s + f \dot{\theta}_p + c \dot{\theta}_e)^2
 \end{aligned} \tag{21}$$

and

$$\begin{aligned}
 Q_{s\varphi} &= \frac{1}{2} \rho A_s C_s (r_b + a) (r_b \dot{\varphi}_b + a \dot{\varphi}_s)^2 \\
 Q_{p\varphi} &= \frac{1}{2} \rho A_p C_p (r_b + a + f) (r_b \dot{\varphi}_b + a \dot{\varphi}_s + f \dot{\varphi}_p)^2 \\
 Q_{e\varphi} &= \frac{1}{2} \rho A_e C_e (r_b + a + f + c) (r_b \dot{\varphi}_b + a \dot{\varphi}_s + f \dot{\varphi}_p + c \dot{\varphi}_e)^2
 \end{aligned} \tag{22}$$

To properly introduce these terms into the $[B]$ matrix, we define a column damping matrix $[D]$ whose elements can be shown to be

$$\begin{aligned}
 D_1 &= Q_{b\theta} S_{bb} / m_4 r_b^2 & D_5 &= Q_{p\varphi} S_{pp} / m_2 f^2 \\
 D_2 &= Q_{b\varphi} S_{bb} / m_4 r_b^2 \theta_b^2 & D_6 &= Q_{p\varphi} S_{pp} / m_2 f^2 \theta_p^2 \\
 D_3 &= Q_{s\theta} S_{ss} / m_3 a^2 & D_7 &= Q_{e\theta} S_{ee} / m_1 c^2 \\
 D_4 &= Q_{s\varphi} S_{ss} / m_3 a^2 \theta_s^2 & D_8 &= Q_{e\varphi} S_{ee} / m_1 c^2 \theta_e^2
 \end{aligned} \tag{23}$$

where the S parameters are given in the Appendix. Thus, the

inclusion of damping terms can be accomplished by adding the damping matrix $[D]$ to $[B]$, i.e.,

$$[B]_{\text{damped}} = [B]_{\text{undamped}} + [D] \quad (24)$$

Precessional Motion

Precessional motion in this paper is modeled after the standard Foucault pendulum analysis found in many classical mechanics texts (e.g., Marion¹⁵, Symon¹⁶, Becker¹⁷, or Goldstein¹⁸). In this framework, we consider motion for the case where $\omega_y = 0$ and $\dot{z} = 0$ (which is consistent with our neglect of balloon bounce).

The Coriolis force $\vec{F}_{\text{cor}} = -2m(\vec{\omega} \times \vec{v})$, when applied through a distance \vec{r} , produces an amount of work $W = 2m \vec{v} \cdot (\vec{\omega} \times \vec{r})$, which may also be written in the Levi-Civita density formalism as

$$W = 2m \sum_{ijk} \epsilon_{ijk} \dot{r}_i \omega_j r_k \quad (25)$$

We consider Equation (25) to be the total energy due to Coriolis forces, so that, in Cartesian coordinates, the Coriolis contribution to the Lagrangian is

$$L_{\text{cor}} = -2m[\dot{x}\omega_z y + \dot{y}(\omega_x z - \omega_z x)] \quad (26)$$

Transforming to spherical coordinates and using the small-angle approximation yields the precessional EOM^{14,15}

$$\begin{aligned} \ddot{\theta}_b &= 2m_b \omega_o r_b^2 [2(\xi_b - \mu_b \varphi_b) \sin \lambda + \xi_b \theta_b \cos \lambda] \\ \ddot{\varphi}_b &= 2m_b \omega_o r_b^2 [-2\theta_b (\xi_b \varphi_b + \mu_b) \sin \lambda] \end{aligned} \quad (27)$$

$$\begin{aligned} \ddot{\theta}_s &= 2(m_b + m_s) a \omega_o \left\{ 2[(r_b \xi_b + a \xi_s) - (r_b \mu_b + a \mu_s) \varphi_s] \sin \lambda \right. \\ &\quad \left. + \theta_s (r_b \xi_b + a \xi_s) \cos \lambda \right\} \end{aligned} \quad (28)$$

$$\ddot{\varphi}_s = 2(m_b + m_s) a \omega_o \left\{ -2\theta_s [(r_b \xi_b + a \xi_s) \varphi_s + (r_b \mu_b + a \mu_s)] \sin \lambda \right\}$$

$$\begin{aligned} \ddot{\theta}_p &= 2(m_b + m_s + m_p) f \omega_o \left\{ 2[(r_b \xi_b + a \xi_s + f \xi_p) - (r_b \mu_b + a \mu_s + f \mu_p) \varphi_p] \sin \lambda \right. \\ &\quad \left. + \theta_p (r_b \xi_b + a \xi_s + f \xi_p) \cos \lambda \right\} \end{aligned} \quad (29)$$

$$\begin{aligned}
\ddot{\varphi}_p &= 2(m_b+m_s+m_p)f\omega_o \left\{ -2\theta_p [(r_b\xi_b+a\xi_s+f\xi_p)\varphi_p+(r_b\mu_b+a\mu_s+f\mu_p)] \sin \lambda \right. \\
\ddot{\theta}_e &= 2(m_b+m_s+m_p+m_e)c\omega_o \left\{ 2[(r_b\xi_b+a\xi_s+f\xi_p+c\xi_e) \right. \\
&\quad \left. -(r_b\mu_b+a\mu_s+f\mu_p+c\mu_e)\varphi_e] \sin \lambda \right. \\
&\quad \left. +\theta_e(r_b\xi_b+a\xi_s+f\xi_p+c\xi_e) \cos \lambda \right\} \\
\ddot{\varphi}_e &= 2(m_b+m_s+m_p+m_e)c\omega_o \left\{ -2\theta_e [(r_b\xi_b+a\xi_s+f\xi_p+c\xi_e)\varphi_e \right. \\
&\quad \left. +(r_b\mu_b+a\mu_s+f\mu_p+c\mu_e)] \sin \lambda \right\}
\end{aligned} \tag{30}$$

where for convenience and brevity we use the abbreviations

$$\mu_k = \dot{\theta}_k - \dot{\varphi}_k \theta_k \varphi_k \qquad \xi_k = \dot{\theta}_k \varphi_k + \dot{\varphi}_k \theta_k \qquad k=b,s,p,e$$

As with the torsional EOM, the precessional EOM exhibit only stat-coupling and may be integrated directly. They are added to the pendulation EOM in a fashion similar to that done in the torsional case; likewise, an analytic solution is possible but prohibitively cumbersome. Precessional motion is not expected to affect the motion of the system significantly; this will be discussed later.

GUST LOADING MODELING

The gust loading model may properly be considered to be the most important aspect of the problem, since the character of the gust field will determine the initial conditions(IC) for solution of the EOM. We will try to define the IC in terms of the gust velocity profile incident on the system. Statistical variation in the gust field or gust effects on supporting cables will be neglected, since they are not expected to be significant in a small-angle deflection analysis.

Gust loading is expected to occur suddenly and to last for a specified finite time interval before being "turned off". If this time interval is small compared to the time interval in which the system is in motion after being loaded, the gust model may be developed as an impulsive motion problem.¹⁴ Denote the time of onset of the gust by t_1 and its "turn-off" time by t_2 . The linear impulse is defined by

$$P_{\sigma} = \Delta(mv) = m(v_2 - v_1) = \int_{t_1}^{t_2} F dt \quad (31)$$

and the applied torque by $N = \dot{L} = I\dot{\omega}$. Hence the impulsive torque J is

$$J = \int_{t_1}^{t_2} N dt = I(\omega_2 - \omega_1) = I \Delta \omega \quad (32)$$

If the primary linear impulse P_{σ} is applied such that its line of action is at distance \mathcal{J} from the CM of the element in question, then $N = F\mathcal{J}$ and hence

$$J = \mathcal{J} \int_{t_1}^{t_2} F dt = P_{\sigma} \mathcal{J} \quad (33)$$

Equating (32) and (33), the angular velocity change produced by a linear impulse acting at distance \mathcal{J} from the CM is $\Delta \omega = P_{\sigma} \mathcal{J} / I$. If a number n of different impulses $P_{\sigma i}$ are applied simultaneously at different distances \mathcal{J}_i , then, noting that the $P_{\sigma i}$ and hence also $\Delta \omega$, are now functions of \mathcal{J}_i as well as of time, we have in general

$$\Delta \omega (\mathcal{J}_i) = \frac{1}{I} \sum_{i=1}^n P_{\sigma i} (\mathcal{J}_i) \mathcal{J}_i \quad (34)$$

which gives the differential change in angular velocity produced during the time interval in which the gust acts. We are primarily interested in motion during a specified time interval after the gust loading has occurred. Since damping forces will influence the motion of the system after gust loading, then we may expect $\Delta \omega$ to be diminished by some function of time (perhaps an exponential decay type behavior) $f(t)$ for $t > 0$ due to damping. The actual temporal-spatial inter-relationships are not clear at this time and will require further study. Thus, for $t > t_2$ we modify Equation (34) to read

$$\Delta \omega (\mathcal{J}_i) = \frac{1}{I} \sum_{i=1}^n P_{\sigma i} (\mathcal{J}_i) \mathcal{J}_i f(t) \quad (35)$$

We also note that translational motion may also suffer the effect of damping.

Spatial Character of Gust Field

Deferring for the present the question of the temporal factor, let us consider the effects of the spatial character of the gust field upon the system. Since we have assumed spherical symmetry for the balloon and cylindrical symmetry for the other system elements, on velocity field profiles that are asymmetric with respect to the element CM will yield a net rotational motion on that element. The geometry of the problem for the balloon is shown in Fig.2 for a profile whose peak is offset from the CM, and similar geometries exist for the other system elements.

For the moment, confine the analysis to a planar velocity profile parallel to the azimuth plane. Then

$$P_{\sigma_i}^k(\zeta_i) = m_k [v(\zeta_i) - v(\zeta_{i-1})] \quad (36)$$

and:

$$\Delta\omega_k(\zeta_i) = \frac{1}{I_k} \sum_{i=1}^n P_{\sigma_i}^k(\zeta_i) \zeta_i f(t) \quad (37)$$

where: $k = b, s, p$ or e . Substituting (36) into (40),

$$\Delta\omega_k(\zeta_i) = \frac{m_k}{I_k} \sum_{i=1}^n [v(\zeta_i) - v(\zeta_{i-1})] \zeta_i f(t) \quad (38)$$

Writing out the first few terms of (38) will show that, upon adding up all terms, we have

$$\omega_k(\zeta_n) = \omega_k(\zeta_0) + \frac{m_k}{I_k} \sum_{i=1}^n (n-i+1) \zeta_i [v(\zeta_i) - v(\zeta_{i-1})] f(t) \quad (39)$$

By Fig.2, ζ_0 is located at the CM (also the geometric center), so that no net rotational impulse is produced at that point; only translational motion could be produced at ζ_0 . Thus, in all cases, $\omega(\zeta_0) = 0$ so that (39) becomes:

$$\omega_k(\zeta_n) = \frac{m_k}{I_k} \sum_{i=1}^n (n-i+1) \zeta_i [v(\zeta_i) - v(\zeta_{i-1})] f(t) \quad (40)$$

Equation (40) thus gives the total accumulated ω out to distance $\zeta = \zeta_n$, but due to the symmetry of the system, this is only half

of the total ω which must also include the contributions from $\zeta = 0$ to $\zeta = -\zeta_n$. Hence we must sum over $2n$ points and (40) now reads

$$\Omega_k = \omega_k(\pm \zeta_n) = \frac{m_k}{I_k} \sum_{i=1}^{2n} (n-i+1) \zeta_i [v(\zeta_i) - v(\zeta_{i-1})] f(t) \quad (41)$$

This expression provides the IC for the problem by defining the net angular velocity change in terms of incident gust velocity profile.

Two- and Three-Dimensional Gust Configurations

The analysis above assumed a planar velocity profile parallel to the azimuth plane. We might also consider planar profiles rotated about the azimuth plane, three-dimensional profiles normal to the z-axis, or more generally, planar and three-dimensional profiles incident on the system at angles off the horizontal (e.g., a gust coming up from below at some specified angle). Consider each case separately, assuming the system at rest prior to load.

We first study a planar profile rotated about the azimuth plane by an angle χ . Assume the characteristic length of the element to be h_k and the point of intersection of the profile at $\zeta = 0$ and the system element occurring at distance h_k^i from the CM along the body axis. In keeping with the coordinate system definition, $h_k^i > 0$ below the CM. At angle χ , the body will see a ζ different from what it saw in the azimuth plane, i.e.,

$$\zeta^i = \zeta \sec \chi \quad (42)$$

This equation, however, becomes singular as $\chi \rightarrow \pm \pi/2$. But we are concerned only with the effect of the gust over h_k . Denote $\zeta > 0$ by ζ_+ and $\zeta < 0$ by ζ_- . Then from Fig. 3, we find that as $\chi \rightarrow \pm \pi/2$ the body effectively sees:

$$\begin{aligned} \zeta_+^i &\rightarrow h_k^i + \frac{1}{2}h_k & , & \quad \zeta_-^i \rightarrow \frac{1}{2}h_k - h_k^i & , & \quad \left(\begin{matrix} 90^\circ \\ 270^\circ \end{matrix} \right) < \chi < \left(\begin{matrix} 180^\circ \\ 360^\circ \end{matrix} \right) \\ \zeta_+^i &\rightarrow \frac{1}{2}h_k - h_k^i & , & \quad \zeta_-^i \rightarrow h_k^i + \frac{1}{2}h_k & , & \quad \left(\begin{matrix} 0^\circ \\ 180^\circ \end{matrix} \right) < \chi < \left(\begin{matrix} 90^\circ \\ 270^\circ \end{matrix} \right) \end{aligned} \quad (43)$$

Equations (42) and (43) provide further generalization of (41),

as we can now specify Ω for any planar profile rotated about the azimuth plane, observing the restrictions of Equation (43); the cases $\chi = 0$ and $\chi = \pi$ are included in this generality due to (42).

Now extend the analysis to a three-dimensional profile normal to the z-axis (Fig. 4). Such a profile may consist of a series of identical parallel planar profiles "stacked" one upon another and either parallel to or rotated about the azimuth plane, or a profile whose volume shape is that of a body of revolution. The geometry for a planar stacked profile is that shown in Fig. 4, and is an extension of the profile shape used in the derivation of Equation (41). For the stacked profile rotated about the azimuth plane (Fig. 5), we assume that there exist a number m of planar profiles parallel to each other rotated about the azimuth plane at angle χ . For each planar the profile, the element will see a different $\zeta_j^!$ and hence will experience a different net angular velocity change Ω_j . The total Ω_k produced on each element is then

$$\Omega_k^{\text{fin}} = \frac{1}{m} \sum_{j=1}^m \Omega_{kj} \quad (44)$$

observing the restrictions of (43), and with (42) modified to

$$\zeta_j^! = \zeta_j \sec \chi \quad (45)$$

For the case of a profile whose shape is that of a body of revolution, regard the profile as a number q of planar parallel stacked profiles, each parallel to the azimuth plane (Fig. 6). It is clear that the spatial character of each such profile is dependent on the distance z from the spherical coordinate system origin, i.e., we have $v(\zeta_i) \rightarrow v(\zeta_i, z)$. The effective velocity profile then becomes

$$v_e(\zeta_i, z) = \frac{1}{q} \sum_{i=1}^q v_i(\zeta_i, z) \quad (46)$$

so that Equation (41) should now read

$$\Omega_k^q = \frac{m_k}{I_k} \sum_{i=1}^{2n} (n-i+1) \zeta_i [v(\zeta_i, z) - v(\zeta_{i-1}, z)] f(t) \quad (47)$$

In general, it is possible for any gust to approach the balloon system at some angle ψ relative to the horizontal (Fig. 6). For a given gust configuration, the variable under study (denoted arbitrarily here by A) will have a horizontal component $A \cos \psi$, and a vertical component $A \sin \psi$. These components must be taken into account and defined before proceeding with a given computation. For this study, we would be interested in the horizontal component.

We have said nothing about the definition of the velocity profile itself. For the LACATE mission, Davis¹⁹ indicates that a gust may load the system for a time interval of 6 seconds and have a peak velocity of 4.0 knots. The time history of this load would be a step function, and the spatial shape is expected to be a $(1 - \cos^2 \epsilon)$ type waveform spread over a distance of about 300 meters. Thus, we would define our velocity profile by

$$v(\xi_i) = \begin{cases} \frac{1}{2} v_{\text{peak}} \left[1 - \cos^2 \frac{\pi}{\lambda} (-\xi_i - \nu - \lambda) \right] & , \quad \xi < 0 \\ \frac{1}{2} v_{\text{peak}} \left[1 - \cos^2 \frac{\pi}{\lambda} (\xi_i - \nu - \lambda) \right] & , \quad \xi > 0 \end{cases} \quad (48)$$

where ν is the distance ξ at which the peak of the profile occurs and λ is the profile quarter-wavelength.

Indeed, other balloon missions may use other velocity profiles. They may also anticipate gust field configurations other than those discussed in this paper, but most of them will probably be reducible to one of the cases discussed here. As more knowledge of the nature of gust fields that occur in reality is gained, it will be possible to make this gust loading model more general and better able to describe the gust as it occurs in nature.

CONCLUSIONS

We will deal with no lengthy conclusions in this report, as it is felt that such conclusions should be backed up with more in-depth study, particularly with respect to the stability criteria for the system. Therefore, we will defer such discussion until the follow-up study on the stability criteria has been completed. Here, we will

make only passing mention of certain observations that may be made from the study and computer program results.

The motion of the system is extremely complex, as we might intuitively expect. This is definitely borne out by a number of test cases run on the computer program. As we noted in the analysis section of this report, strong static and dynamic coupling between the various elements of the system is evidenced by the computer program. We also find that the contributions due to the Coriolis effect are very small indeed, as we would expect. Damping effects appear to be negligible for approximately the first three minutes in any given test case; after this time, these effects gradually begin to be more significant, although they do not appear to become an overriding or even major factor in the overall system motion. However, when we consider torsional motion alone, the damping effects do appear to become a significant part of the motion after a few minutes.

The most significant type of motion appears to be the pendulatory motion. The magnitude of this motion varies from component to component of the balloon system. The balloon exhibits, in a typically convergent set of solutions, the most pronounced degree of pendulation. This would be expected from our current model, since we have assumed that the gust is applied spatially only over the dimensions of the balloon; subsequent motion of the other system components is assumed to be induced by the motion of the balloon. We find that the balloon may take rather large excursions in both the θ and φ directions; in the same test case, the load bar made major excursions only in the φ direction, although to a lesser degree than did the parachute package. The payload made only small excursions in the θ direction; this fact is further backed up by preliminary data from the actual LACATE flight, which indicated very little motion in the payload package.

It may also be noted that for the case of simple damped pendulatory motion in a plane, the analytical results obtained in this study reduce to those obtained by Hinton¹⁰.

Although we have been remiss in making any really concrete conclusions in this report, we feel that more accurate and meaningful conclusions can be drawn after further study into the stability question, and after LACATE flight data are available to verify the correctness of the model proposed here.

COMPUTER PROGRAM STRUCTURE AND USAGE

Program LACATE consists of the driver program LACATE and attendant subroutines GUST, DIFSUB, PEDERV, MATINV, DIFFUN, PEND, TORS, CORL, DAMP, BOSS and PLOTS. The basic program structure is shown in Figure 7. The program has been modularized by function to facilitate its use so that only those modules that are desired for a particular run are executed. The gust model module, which provides the $\dot{\theta}_b$ and $\dot{\psi}_b$ initial conditions, constitutes subroutine GUST. Subroutine DIFSUB is the main integrating module, and has subroutine DIFFUN as an externally defined argument. DIFFUN is used to define the derivatives of the equations of motion to be integrated. It contains logic used to call up the pendulation module PEND, the torsion module TORS, and/or the Coriolis module CORL as desired. The module PEND also has a logical flag that may be set to call the damping module DAMP if desired. A short discussion of each module follows; for more detailed information, the user should consult the program listing documentation.

Under the current program configuration, motion of the balloon simulated over a 30 second time interval requires approximately 60 seconds of CPU time to compile and execute; it requires 45K₈ locations of core and expends 400 O/S calls, and is designed for use on the CDC 6600.

LACATE

This is the mainline program which reads all NAMELIST inputs, calls the gust module to set up initial conditions, sets up the inputs for DIFSUB, and makes the call to DIFSUB, which integrates the equations of motion. This routine also controls all printed, punched and plotted output functions.

GUST

This subroutine computes the initial conditions for the angular velocities of the balloon. It requires as inputs the following variables read from cards:

ALPHA - the quarter wavelength of the sinusoidally shaped gust velocity profile.

NU - the distance \mathcal{L} at which the peak of the gust velocity profile occurs.

NRB - number of \mathcal{L} points to be used.

It provides as output the values TDOTB and DOTB, which are the initial conditions on $\dot{\theta}_b$ and $\dot{\varphi}_b$, respectively.

DIFSUB

This subroutine integrates the system of first order differential equations of motion. In our case, we have a "system of systems" of ordinary differential equations. It requires as inputs the data in NAMELIST /DIFL/. The equations to be integrated are defined and set up in DIFFUN.

DIFFUN

Subroutine DIFFUN computes the scaled derivatives and sets up the equations of motion to be integrated by DIFSUB. By setting the logical flags appropriately, the user may choose any desired combination of the modules PEND, TORS and CORL. After this has been done, the total contribution from each module is added up to give a cumulative total due to the combinations before returning to DIFSUB for integration. If a module is not used, its contribution is automatically preset to zero.

PEND

Subroutine PEND sets up the pendulatory equations of motion. The damping module DAMP may be called, if desired, to add in the contributions of viscous and aerodynamic damping.

TORS

This subroutine sets up the equations of motion for plane torsional motion. This module already contains its own damping terms.

CORL

Subroutine CORL sets up the equations of motion for inclusion of the Coriolis effect on the motion of the system.

DAMP

Subroutine DAMP computes the viscous and aerodynamic damping terms to be added into the pendulation module. The damping terms are passed back to PEND via the D(I) array and added to the B(I) array in PEND.

BOSS

Subroutine BOSS is a SNOBOL 4.0 generated routine which solves a system of linear equations represented by the matrix equation $[A][\ddot{q}] = [B]$. BOSS is utilized only by subroutine PEND to solve the system of eight linear equations of motion. The algorithm employed is the method of detached coefficients, alternatively called the method of Gaussian elimination.

PLOTS

This routine calls up the Calcomp plotting routines TPLOT and AVPLOT.

Sample inputs, outputs and a program listing are given in Appendix B.

PLOTTING ROUTINES FOR LACATE

Two post-processor plotting routines have been written to provide plots for program LACATE. These are programs TPLOT and AVPLOT.

PROGRAM TPLOT

This program uses the punched output of the angles θ and φ for each component of the balloon system from program LACATE to provide a plot of THETA versus PHI. A point is plotted after each pass through DIFSUB, so the user may observe the spatial motion of each component as seen from above that component. All four plots (balloon, load bar, parachute package and payload) are plotted together, with the same scaling factors, to provide a comparative view of the magnitude of each component's motion relative to the other components.

The program is written for the Calcomp plotter, and requires as inputs the punched output of the angles from DIFSUB. The number of passes through DIFSUB, and hence the number of punched

cards, is denoted by the variable NB, which is set by the user. The scaling may also be reset by the user by changing the adjusted minimum AMIN and scale factor ASCF as dictated by the range of data being dealt with; this will allow the user to use as much space on the plotting paper as possible in order to obtain the best detail.

The program may be run successfully in 50 seconds, 50K₈ core and 700 O/S calls; No. 301 Red plotting paper is recommended.

PROGRAM AVPLOT

Program AVPLOT is a Calcomp routine intended to provide plots of the solutions of the equations of motion as a function of time. It requires as inputs the punched card output from LACATE for the angular velocities as well as the angles. At this writing, however, AVPLOT has not been completely debugged and tested, and thus further discussion of it will be deferred until future updates on LACATE.

CONCLUDING REMARKS

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APPENDIX A

ELEMENTS OF THE DYNAMIC AND STATIC COUPLING COEFFICIENT MATRICES:

For brevity in writing out the terms of the [A] and [B] matrices, we first introduce the following expressions:

$$\begin{aligned}
 S_b &= (I_b/m_4 r_b^2 + 1)^{-1} & S_{bb} &= (I_b/m_4 r_b \theta_b^2 + 1)^{-1} \\
 S_s &= (I_s/m_3 a^2 + 1)^{-1} & S_{ss} &= (I_s/m_3 a \theta_s^2 + 1)^{-1} \\
 S_p &= (I_p/m_2 f^2 + 1)^{-1} & S_{pp} &= (I_p/m_2 f \theta_p^2 + 1)^{-1} \\
 S_e &= (I_e/m_1 c^2 + 1)^{-1} & S_{ee} &= (I_e/m_1 c \theta_e^2 + 1)^{-1}
 \end{aligned}$$

The elements of the dynamic coupling coefficient matrix [A] are then written

$$A_{12}=A_{14}=A_{21}=A_{32}=A_{34}=A_{43}=A_{56}=A_{58}=A_{65}=A_{76}=A_{78}=A_{87} = 0$$

$$A_{11}=A_{22}=A_{33}=A_{44}=A_{55}=A_{66}=A_{77}=A_{88} = 1$$

$$A_{13} = m_3 a S_b / m_4 r_b$$

$$A_{15} = m_2 f S_b / m_4 r_b$$

$$A_{16} = m_2 f_p (\varphi_b - \varphi_p) S_b / m_4 r_b$$

$$A_{17} = m_1 c S_b / m_4 r_b$$

$$A_{18} = m_1 c \theta_e (\varphi_b - \varphi_e) S_b / m_4 r_b$$

$$A_{23} = A_{24} = m_3 a \varphi_s S_{bb} / m_4 r_b \theta_b$$

$$A_{25} = A_{26} = m_2 f \varphi_p S_{bb} / m_4 r_b \theta_b$$

$$A_{27} = A_{28} = m_1 c \varphi_e S_{bb} / m_4 r_b \theta_b$$

$$A_{31} = r_b S_s / a$$

$$A_{35} = m_2 f S_s / m_3 a$$

$$A_{36} = m_2 f \theta_p (\varphi_s - \varphi_p) S_s / m_3 a$$

$$A_{37} = m_1 c S_s / m_3 a$$

$$A_{38} = m_1 c \theta_e (\varphi_s - \varphi_e) S_s / m_3 a$$

$$A_{41} = r_b \varphi_b S_{ss} / a \theta_s = A_{42}$$

$$A_{45} = A_{46} = m_2 f \varphi_p S_{ss} / m_3 a \theta_s$$

$$A_{47} = A_{48} = m_1 c \varphi_e S_{ss} / m_3 a \theta_s$$

$$A_{51} = r_b S_p / f$$

$$A_{52} = r_b \theta_b (\varphi_p - \varphi_b) S_p / f$$

$$A_{53} = a S_p / f$$

$$A_{54} = a \theta_s (\varphi_p - \varphi_s) S_p / f$$

$$A_{57} = m_1 c S_p / m_2 f$$

$$A_{61} = A_{62} = r_b \varphi_b S_{pp} / f \theta_p$$

$$A_{63} = A_{64} = a \varphi_s S_{pp} / f \theta_p$$

$$A_{67} = A_{68} = m_1 c \varphi_e S_{pp} / m_2 f \theta_p$$

$$A_{71} = r_b S_e / c$$

$$A_{72} = r_b \theta_b (\varphi_e - \varphi_b) S_e / c$$

$$A_{73} = a S_e / c$$

$$A_{74} = a \theta_s (\varphi_e - \varphi_s) S_e / c$$

$$A_{75} = f S_e / c$$

$$A_{81} = A_{82} = r_b \varphi_b S_{ee} / c \theta_e$$

$$A_{83} = A_{84} = a\dot{\varphi}_s S_{ee}/c \dot{\theta}_e$$

$$A_{85} = A_{86} = f\dot{\varphi}_p S_{ee}/c \dot{\theta}_e$$

and the elements of the column static coupling coefficient matrix [B] are written

$$B_1 = S_b \left\{ \dot{\varphi}_b^2 \theta_b - m_3 (-a\dot{\varphi}_s^2 \theta_s + g\dot{\theta}_b) / m_4 r_b \right. \\ \left. - m_2 f [2\dot{\theta}_p \dot{\varphi}_p (\varphi_b - \varphi_p) - \dot{\varphi}_p^2 \theta_p + \dot{\theta}_p^2 (\theta_b - \theta_p)] / m_4 r_b \right. \\ \left. - m_1 c [2\dot{\theta}_e \dot{\varphi}_e (\varphi_b - \varphi_e) - \dot{\varphi}_e^2 \theta_e + \dot{\theta}_e^2 (\theta_b - \theta_e)] / m_4 r_b \right\}$$

$$B_2 = S_{bb} \left\{ -2(m_3 a \dot{\theta}_s \dot{\varphi}_s + m_2 f \dot{\theta}_p \dot{\varphi}_p + m_1 c \dot{\theta}_e \dot{\varphi}_e) / m_4 r_b \theta_b - 2\dot{\theta}_b \dot{\varphi}_b / \theta_b \right\}$$

$$B_3 = S_s \left\{ \dot{\varphi}_s^2 \theta_s - (-r_b \dot{\varphi}_b^2 \theta_b + g\dot{\theta}_s) / a \right. \\ \left. - m_2 f [2\dot{\theta}_p \dot{\varphi}_p (\varphi_s - \varphi_p) - \dot{\varphi}_p^2 \theta_p + \dot{\theta}_p^2 (\theta_s - \theta_p)] / m_3 a \right. \\ \left. - m_1 c [2\dot{\theta}_e \dot{\varphi}_e (\varphi_s - \varphi_e) - \dot{\varphi}_e^2 \theta_e + \dot{\theta}_e^2 (\theta_s - \theta_e)] / m_3 a \right\}$$

$$B_4 = S_{ss} \left\{ -2(m_3 r_b \dot{\theta}_b \dot{\varphi}_b + m_2 f \dot{\theta}_p \dot{\varphi}_p + m_1 c \dot{\theta}_e \dot{\varphi}_e) / m_3 a \theta_s - 2\dot{\theta}_s \dot{\varphi}_s / \theta_s \right\}$$

$$B_5 = S_p \left\{ \dot{\varphi}_p^2 \theta_p - r_b [-\dot{\varphi}_b^2 \theta_b + \dot{\theta}_b^2 (\theta_p - \theta_b) + 2\dot{\theta}_b \dot{\varphi}_b (\varphi_p - \varphi_b)] / f \right. \\ \left. - a [-\dot{\varphi}_s^2 \theta_s + \dot{\theta}_s^2 (\theta_p - \theta_s) + 2\dot{\theta}_s \dot{\varphi}_s (\varphi_p - \varphi_s)] / f \right. \\ \left. + m_1 c \dot{\varphi}_e^2 \theta_e / m_2 f + g\dot{\theta}_p / f \right\}$$

$$B_6 = S_{pp} \left\{ -2[m_2 (r_b \dot{\theta}_b \dot{\varphi}_b + a\dot{\theta}_s \dot{\varphi}_s) + m_1 c \dot{\theta}_e \dot{\varphi}_e] / m_2 f \theta_p - 2\dot{\theta}_p \dot{\varphi}_p / \theta_p \right\}$$

$$B_7 = S_e \left\{ \dot{\varphi}_e^2 \theta_e - r_b [-\dot{\varphi}_b^2 \theta_b + \dot{\theta}_b^2 (\theta_e - \theta_b) + 2\dot{\theta}_b \dot{\varphi}_b (\varphi_e - \varphi_b)] / c \right. \\ \left. - a [-\dot{\varphi}_s^2 \theta_s + \dot{\theta}_s^2 (\theta_e - \theta_s) + 2\dot{\theta}_s \dot{\varphi}_s (\varphi_e - \varphi_s)] / c \right. \\ \left. - f\dot{\varphi}_p^2 \theta_p / c + g\dot{\theta}_e / c \right\}$$

and

$$B_8 = S_{ee} \left\{ -2(r_b \dot{\theta}_b \dot{\varphi}_b + a\dot{\theta}_s \dot{\varphi}_s + f\dot{\theta}_p \dot{\varphi}_p) / c \theta_e - 2\dot{\theta}_e \dot{\varphi}_e / \theta_e \right\}$$

APPENDIX B

SAMPLE INPUTS AND OUTPUTS AND LISTING OF PROGRAM LACATE

This appendix provides data inputs and output for a typical 30 second run, as well as a complete listing of the LACATE program.

Inputs

All inputs are made through the use of NAMELIST statements set up according to their functional use. The following tables describe these inputs in more detail; unless a value is otherwise specified, a particular variable is to be set by the user. All inputs are made in the MKS system of units.

NAMELIST	VARIABLES/CONSTANTS	FUNCTION
ANGL	THETAB, THETAS, THETAP, THETAE, PHIB, PHIS, PHIP, PHIE, TDOTS, TDOTP, DOTE, DOTS, DOTP, DOTE, OMEGA, LAMDA	Inputs initial conditions on angles, angular velocities and Coriolis constants
CONS	MB, MS, MP, ME, RB, AL, L, P, RS, RP, RE, G, RHO, MU, N, HS, HP, HE	Inputs constants used in program
TEST	IPEND, ITORS, ICORL, IPLOTS, IDAMP	Inputs logical flags for module selection
DIF1	H, JSTART, HMIN, HMAX, EPS, MTH, MAXDER, TO, TF, PV	Initialization constants for DIFSUB
DRAG	ASCS, APCP, AECE	Constants used in aerodynamic damping computations

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Outputs

Program LACATE provides both printed and punched output. Printer output consists of the THETA and PHI angles and their corresponding angular velocities for each component of the system, i.e., the solutions of the equations of motion. These values are printed after each successful pass through DIFSUB at intervals of time determined by the step size chosen by DIFSUB. Printed output is also punched on cards to be used as inputs to the two post-processor plotting routines. Output from a typical test case follow. For each time indicated, the top line gives the values of the angles, while directly below each angle value is given its corresponding angular velocity at that time. The case shown was run for TF = 30.0 seconds. At the end of this time, DIFSUB was still providing convergent solutions and a normal termination occurs. In the case of a non-convergent solution, output would continue as in the convergent case, but when the minimum step size specified by DIFSUB is reached, program execution is halted and an error code and working storage array are dumped for diagnostic purposes. The angles have units of radians, and the angular velocities are given in radians/sec.

Also shown are representative gust model computations for the velocity profile (A) of Equation (48) in the text. Referring to Figure , we interpret this data as follows:

- I - the value of ζ , i.e., the distance outward from the center of the balloon.
- AA(I) - the velocity at station I for $\zeta < 0$.
- BB(I) - the velocity at station I for $\zeta > 0$.
- V(J) - the velocity at station I considering both $\zeta < 0$ and $\zeta > 0$, i.e., $V(I) = AA(I) + BB(I)$. These values of V are used to compute the rotational impulse at station I due to $V(I)$.

The computation of the net rotational impulse produced by the gust at each station I, i.e., the computation of Ω , is also shown. Note that as one proceeds from I=1 to I=150, the net impulse is accumulating. In other words, $W(I)$ is the net change in angular velocity that has been produced by all velocity components from I=1 to I=I(current). Thus the final value of $W(I)$ (here, $W(150)$) is the net change in angular velocity over the entire balloon, and it is this value that is taken as the initial condition(s) for $\dot{\theta}_b$ and/or $\dot{\psi}_b$.

PARAMETER	VALUE	DESCRIPTION
THETAB	radians	balloon theta deflection
THETAS	radians	load bar theta deflection
THETAP	radians	parachute package theta deflection
THETA E	radians	payload theta deflection
PHIB	radians	balloon phi deflection
PHIS	radians	load bar phi deflection
PHIP	radians	parachute phi deflection
PHIE	radians	payload phi deflection
TDOTS	rad/sec	load bar angular velocity in theta direction
TDOTP	rad/sec	parachute package angular velocity, theta direction
TDOTE	rad/sec	payload angular velocity, theta direction
DOTS	rad/sec	load bar angular velocity, phi direction
DOTP	rad/sec	parachute package angular velocity, phi direction
DOTE	rad/sec	payload angular velocity, phi direction
OMEGA	$7.27 \times 10^{-5} \frac{\text{rad}}{\text{sec}}$	rotational velocity of the earth; used only in CORL
LAMDA	$5.7305 \times 10^{-1} \text{rad}$	geometric latitude at Holloman Air Force Base, New Mexico; used only in CORL
MB	861.82 kg.	mass of balloon
MS	112.94 kg.	mass of load bar
MP	37.194 kg.	mass of parachute package
ME	214.095 kg.	mass of payload package
M	M(1)=214.095 kg. M(2)=251.29 M(3)=364.23 M(4)=1226.06	An array dimensioned M(4) such that: M(1)=ME M(2)=ME+MP M(3)=ME+MP+MS M(4)=ME+MP+MS+MB
RB	64.00 meters	balloon radius
A1	23.16 meters	length of cable between balloon attachment point and load bar
L	396.24 meters	length of cable between load bar and parachute package
P	3.3527 meters	length of cable between parachute package and payload package

PARAMETER	VALUE	DESCRIPTION
RS	0.6096 meters	radius of load bar cylinder
RP	0.4572 meters	radius of parachute cylinder
RE	0.9244 meters	radius of payload cylinder
G	9.656 m/sec	acceleration due to gravity at an altitude of 50 km;resettable
RHO	$1.027 \times 10^{-3} \frac{\text{kg}}{\text{m}^3}$	density of air at 50 km altitude; user resettable
MU	$1.695 \times 10^{-5} \frac{\text{kg}}{\text{m-sec}}$	viscosity of air at 50 km altitude ; user resettable
N		number of differential equations to be solved =4 for TORS =8 for PEND and CORL
HS	0.6096 meters	length of load bar
HP	0.9144 meters	length of parachute package
HE	1.8288 meters	length of payload package
IPEND		pendulation module flag =0, module not used =1, module is used
ITORS		torsion module flag
ICORL		Coriolis module flag
IPLT		plotting module flag
IDAMP		damping module flag
H, JSTART, HMIN, HMAX, EPS, MTH, MAXDER, TO, TF, PV		refer to subroutine DIFSUB
ASCS	1.03085 m ²	product of load bar surface area and drag coefficient
APCP	1.03085 m ²	same for parachute package
AECE	1.50503 m ²	same for payload package

\$ANGL

THETAB = 0.1E-05.

THEIAS = 0.1E-05.

THETAP = 0.1E-05.

THETAE = 0.1E-05.

PHIB = 0.1E-05.

PHIS = 0.1E-05.

PHIP = 0.1E-05.

PHIE = 0.1E-05.

TDDIS = 0.0.

TDDTP = 0.0.

YDOTE = 0.0.

DOTS = 0.0.

DDIP = 0.0.

DDT = 0.0.

OMEGA = 0.727E-04.

LAMDA = 0.57305E+00.

\$END

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SCONS

MR = 0.86182556E+03.
MS = 0.1129445076E+03.
MP = 0.371945768E+02.
ME = 0.2140956128E+03.
M = 0.2140956128E+03. 0.2512901896E+03. 0.3642346972E+03.
0.12260602572E+04.
RB = 0.64E+02.
AI = 0.23164778E+02.
L = 0.3962387E+03.
P = 0.3352789E+01.
RS = 0.6096E+03.
RP = 0.4572E+00.
RE = 0.9144E+00.
G = 0.9656084E+01.
RHO = 0.1027665726E-02.
MI = 0.169496074855E-04.
N = 8.
HS = 0.6096E+00.
HP = 0.9144E+00.
HF = 0.18288E+01.
SFND

STFST

IPEND = 1.

ITORS = 0.

ICORL = 0.

IPLGT = 0.

IOAMP = 0.

SEND

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SDIF1

H = 0.125E-06.

JSTART = 0.

HMIN = 0.125E-09.

HMAX = 0.1E+00.

EPS = 0.125E-03.

MTH = 0.

MAXDER = 4.

TD = 0.0.

IF = 0.3E+02.

PV = 0.5E-01.

SEND

\$DRAG

ASCS = 0.103085F+01.

APCP = 0.103085F+01.

AFCE = 0.150503E+01.

\$FND

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ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
2.302500	-3.1350E-04	2.302500	-2.46535E-04	2.302500	-3.37432E-04	2.302500	-1.41093E-05	2.302500	-2.99758E-05	2.302500	-1.9450E-07	2.302500	-1.0308E-06	2.302500	-1.73883E-06	2.302500	-2.97026E-07
ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
1.302500	-3.55904E-04	1.302500	-2.44535E-04	1.302500	-3.63415E-04	1.302500	-1.72763E-05	1.302500	-3.33115E-05	1.302500	-2.65509E-07	1.302500	-8.31673E-07	1.302500	-2.14928E-06	1.302500	-8.98554E-07
ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
1.502500	-3.80251E-04	1.502500	-2.44240E-04	1.502500	-3.89398E-04	1.502500	-2.07725E-05	1.502500	-3.09330E-05	1.502500	-1.08451E-06	1.502500	-7.68703E-07	1.502500	-2.58348E-06	1.502500	-3.50354E-07
ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
1.502500	-4.51928E-04	1.502500	-2.35281E-04	1.502500	-4.67347E-04	1.502500	-3.30982E-05	1.502500	-4.66616E-06	1.502500	-1.23779E-06	1.502500	-4.34950E-07	1.502500	-4.00443E-06	1.502500	-5.99163E-07
ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
1.702500	-4.28275E-04	1.702500	-2.37777E-04	1.702500	-4.41364E-04	1.702500	-2.87021E-05	1.702500	-4.25929E-05	1.702500	-5.89197E-07	1.702500	-3.51391E-06	1.702500	-4.83198E-06	1.702500	-9.29095E-07
ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
1.802500	-4.04378E-04	1.802500	-2.40153E-04	1.802500	-4.15381E-04	1.802500	-2.45855E-05	1.802500	-4.49145E-07	1.802500	-1.2443E-06	1.802500	-6.88849E-07	1.802500	-3.33923E-06	1.802500	-7.84451E-07
ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
1.902500	-4.75325E-04	1.902500	-2.32670E-04	1.902500	-4.93330E-04	1.902500	-3.77492E-05	1.902500	-4.76961E-05	1.902500	-8.37729E-07	1.902500	-1.62920E-06	1.902500	-4.50719E-06	1.902500	-6.67108E-07
ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
2.102500	-5.21308E-04	2.102500	-2.27113E-04	2.102500	-5.45297E-04	2.102500	-4.7644E-05	2.102500	-5.15488E-05	2.102500	-1.17428E-06	2.102500	-7.28870E-08	2.102500	-5.53235E-06	2.102500	-1.18256E-07
ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
2.202500	-5.43873E-04	2.202500	-2.24172E-04	2.202500	-5.71280E-04	2.202500	-5.29198E-05	2.202500	-1.64174E-06	2.202500	-3.22148E-07	2.202500	-6.04476E-06	2.202500	-8.04476E-06	2.202500	-1.10149E-07
ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
2.302500	-5.66137E-04	2.302500	-2.21126E-04	2.302500	-5.97263E-04	2.302500	-5.82627E-05	2.302500	-1.78850E-06	2.302500	-6.12501E-07	2.302500	-6.54939E-06	2.302500	-4.54939E-06	2.302500	-3.79148E-07
ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
2.402500	-6.88092E-04	2.402500	-2.17979E-04	2.402500	-6.23246E-04	2.402500	-6.36827E-05	2.402500	-1.95638E-06	2.402500	-9.47879E-07	2.402500	-7.03979E-06	2.402500	-4.81293E-06	2.402500	-6.92828E-07
ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
2.502500	-6.99728E-04	2.502500	-2.14732E-04	2.502500	-6.49229E-04	2.502500	-6.91381E-05	2.502500	-2.14561E-06	2.502500	-1.33224E-06	2.502500	-7.50889E-06	2.502500	-4.56361E-06	2.502500	-1.05534E-06
ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
2.602500	-6.31034E-04	2.602500	-2.11387E-04	2.602500	-6.75211E-04	2.602500	-7.45866E-05	2.602500	-2.36027E-06	2.602500	-1.76953E-06	2.602500	-7.94913E-06	2.602500	-4.79491E-06	2.602500	-1.47085E-06
ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
2.702500	-6.52003E-04	2.702500	-2.07947E-04	2.702500	-7.01194E-04	2.702500	-7.99358E-05	2.702500	-2.50431E-06	2.702500	-2.26366E-06	2.702500	-8.35245E-06	2.702500	-4.82590E-06	2.702500	-1.94354E-06
ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
2.802500	-6.65200E-04	2.802500	-2.04947E-04	2.802500	-7.20119E-04	2.802500	-8.29442E-05	2.802500	-2.50442E-06	2.802500	-2.26366E-06	2.802500	-8.35245E-06	2.802500	-4.82590E-06	2.802500	-1.94354E-06

PHIE

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THETAB

TIME

	TIME	THETAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETAE	PHIE
ANGLE	2.902500	-6.92878E-07	-7.53160E-04	-9.04714E-05	3.15034E-06	-3.43753E-06	-9.01424E-06	-3.07665E-06	9.96692E-07
RATES		-2.00785E-04	-2.59829E-04	-5.10285E-05	3.01719E-06	-6.52114E-06	-2.74483E-06	-6.32675E-06	-1.09752E-03
ANGLE	3.002500	-7.12771E-04	-7.79143E-04	-9.54793E-05	3.46524E-06	-4.12453E-06	-9.25506E-06	-3.74485E-06	9.96592E-07
RATES		-1.97066E-04	-2.59828E-04	-4.91521E-05	3.27884E-06	-7.21717E-06	-2.07060E-06	-7.03532E-06	-9.05973E-10
ANGLE	3.102500	-7.32287E-04	-8.05126E-04	-1.00282E-04	3.80633E-06	-4.88278E-06	-9.42393E-06	-4.48566E-06	9.96512E-07
RATES		-1.93257E-04	-2.59828E-04	-4.69329E-05	3.54071E-06	-7.94581E-06	-1.30765E-06	-7.77851E-06	-6.86254E-11
ANGLE	3.202500	-7.51417E-04	-8.31109E-04	-1.04847E-04	4.17351E-06	-5.71543E-06	-9.51211E-06	-5.30240E-06	9.96456E-07
RATES		-1.89359E-04	-2.59828E-04	-4.44047E-05	3.80046E-06	-8.70476E-06	-4.58657E-07	-8.55355E-06	-4.38957E-10
ANGLE	3.302500	-7.70154E-04	-8.57091E-04	-1.09146E-04	4.56644E-06	-6.62537E-06	-9.51123E-06	-6.19810E-06	9.96426E-07
RATES		-1.85372E-04	-2.59828E-04	-4.16061E-05	4.05570E-06	-9.49147E-06	4.71428E-07	-9.35736E-06	-1.65390E-10
ANGLE	3.402500	-7.88437E-04	-8.83074E-04	-1.13153E-04	4.98454E-06	-7.61524E-06	-9.41353E-06	-7.17547E-06	9.96424E-07
RATES		-1.81298E-04	-2.59828E-04	-3.85800E-05	4.30405E-06	-1.03032E-05	1.47514E-06	-1.01867E-05	1.32331E-10
ANGLE	3.502500	-8.06409E-04	-9.09057E-04	-1.16948E-04	5.42702E-06	-8.68739E-06	-9.21215E-06	-8.23688E-06	9.96453E-07
RATES		-1.77136E-04	-2.59827E-04	-3.53726E-05	4.54312E-06	-1.11368E-05	2.54219E-06	-1.10382E-05	4.51186E-10
ANGLE	3.602500	-8.23910E-04	-9.35040E-04	-1.20216E-04	5.89282E-06	-9.84385E-06	-8.90141E-06	-9.38439E-06	9.96515E-07
RATES		-1.72888E-04	-2.59827E-04	-3.20312E-05	4.77062E-06	-1.19893E-05	3.65929E-06	-1.19086E-05	7.87171E-10
ANGLE	3.702500	-8.40982E-04	-9.61022E-04	-1.23246E-04	6.39288E-06	-1.10864E-05	-8.47712E-06	-1.06197E-05	9.96612E-07
RATES		-1.64554E-04	-2.59827E-04	-2.86127E-05	4.99433E-06	-1.28574E-05	4.80989E-06	-1.27943E-05	1.13523E-09
ANGLE	3.802500	-8.57617E-04	-9.87005E-04	-1.25932E-04	6.88911E-06	-1.24153E-05	-7.93693E-06	-1.19442E-05	9.96743E-07
RATES		-1.64135E-04	-2.59827E-04	-2.51635E-05	5.18217E-06	-1.37378E-05	5.97412E-06	-1.36922E-05	1.48923E-09
ANGLE	3.902500	-8.73805E-04	-1.01299E-03	-1.28275E-04	7.41643E-06	-1.38347E-05	-7.28062E-06	-1.33589E-05	9.96910E-07
RATES		-1.59631E-04	-2.59826E-04	-2.17383E-05	5.36224E-06	-1.46270E-05	7.12873E-06	-1.45987E-05	1.84190E-09
ANGLE	4.002500	-8.89539E-04	-1.03897E-03	-1.30278E-04	7.96077E-06	-1.53423E-05	-6.51050E-06	-1.48645E-05	9.97112E-07
RATES		-1.55043E-04	-2.59826E-04	-1.83894E-05	5.52284E-06	-1.55217E-05	8.24713E-06	-1.55107E-05	2.18491E-09
ANGLE	4.102500	-9.04809E-04	-1.06495E-03	-1.31954E-04	8.52010E-06	-1.69395E-05	-5.63168E-06	-1.64615E-05	9.97347E-07
RATES		-1.50371E-04	-2.59826E-04	-1.51677E-05	5.66248E-06	-1.64186E-05	9.29965E-06	-1.64248E-05	2.50887E-09
ANGLE	4.202500	-9.19609E-04	-1.09094E-03	-1.33316E-04	9.09229E-06	-1.85263E-05	-4.65238E-06	-1.81498E-05	9.97613E-07
RATES		-1.45618E-04	-2.59826E-04	-1.21223E-05	5.77947E-06	-1.73146E-05	1.02538E-05	-1.73377E-05	2.80347E-09
ANGLE	4.302500	-9.33929E-04	-1.11642E-03	-1.34384E-04	9.67505E-06	-2.04025E-05	-3.58422E-06	-1.99291E-05	9.97907E-07
RATES		-1.40782E-04	-2.59826E-04	-9.29923E-06	5.87436E-06	-1.82065E-05	1.10747E-05	-1.82462E-05	3.05765E-09
ANGLE	4.402500	-9.47761E-04	-1.14290E-03	-1.35184E-04	1.02661E-05	-2.22675E-05	-2.44236E-06	-2.17989E-05	9.98223E-07

ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES	ANGLE	RATES
4.502500	-9.61098E-04	-1.6988E-03	-1.35744E-04	1.08629E-05	-2.4220E-05	-1.2470E-06	-2.3758E-05	9.98557E-07	-2.03373E-05	4.602500	-9.73932E-04	-1.19487E-03	-1.36095E-04	1.14632E-05	-2.6260E-05	-1.68807E-08	-2.58060E-05	9.98900E-07	-2.58060E-05
4.802500	-9.9059E-04	-1.24683E-03	-1.36315E-04	1.26645E-05	-3.0595E-05	2.42821E-06	-3.02603E-05	9.99563E-07	-3.02603E-05	4.702500	-9.8654E-04	-1.2205E-04	-1.36273E-04	1.20645E-05	-2.8386E-05	1.21774E-06	-2.79406E-05	9.99246E-07	-2.79406E-05
4.902500	-1.0934E-03	-1.27281E-03	-1.36262E-04	1.32608E-05	-3.2888E-05	3.58125E-06	-3.24633E-05	9.99903E-07	-3.24633E-05	5.002500	-1.0208E-03	-1.36153E-04	-1.36153E-04	1.38512E-05	-3.52607E-05	4.6059E-06	-3.48474E-05	1.00019E-06	-3.48474E-05
5.102500	-1.3029E-03	-1.32478E-03	-1.36031E-04	1.44336E-05	-3.7712E-05	5.56735E-06	-3.73105E-05	1.00044E-06	-3.73105E-05	5.202500	-1.03975E-03	-1.35076E-03	-1.35937E-04	1.50059E-05	-4.02401E-05	6.3202E-06	-3.98504E-05	1.00064E-06	-3.98504E-05
5.202500	-1.03975E-03	-1.35076E-03	-1.35937E-04	1.50059E-05	-4.02401E-05	6.3202E-06	-3.98504E-05	1.00064E-06	-3.98504E-05	5.302500	-9.38847E-05	-2.59824E-04	6.96302E-07	5.66850E-06	-2.5558E-05	6.5506E-06	-2.57769E-05	1.63193E-09	-2.57769E-05
5.302500	-1.06561E-03	-1.42871E-03	-1.36228E-04	1.66438E-05	-4.8269E-05	7.1137E-06	-4.79082E-05	1.00076E-06	-4.79082E-05	5.402500	-1.05762E-03	-1.40273E-03	-1.35997E-04	1.61128E-05	-4.55181E-05	7.13608E-06	-4.51514E-05	1.00081E-06	-4.51514E-05
5.402500	-1.05762E-03	-1.40273E-03	-1.35997E-04	1.61128E-05	-4.55181E-05	7.13608E-06	-4.51514E-05	1.00081E-06	-4.51514E-05	5.502500	-8.83440E-05	-2.59824E-04	-1.35997E-04	1.5563E-05	-4.2842E-05	6.85797E-06	-4.24648E-05	1.00076E-06	-4.24648E-05
5.502500	-1.06561E-03	-1.42871E-03	-1.36228E-04	1.66438E-05	-4.8269E-05	7.1137E-06	-4.79082E-05	1.00076E-06	-4.79082E-05	5.602500	-8.83440E-05	-2.59824E-04	-1.35997E-04	1.5563E-05	-4.2842E-05	6.85797E-06	-4.24648E-05	1.00076E-06	-4.24648E-05
5.602500	-1.07304E-03	-1.45469E-03	-1.36641E-04	1.71577E-05	-5.10784E-05	6.74062E-06	-5.07331E-05	1.00060E-06	-5.07331E-05	5.702500	-1.07989E-03	-1.48067E-03	-1.37268E-04	1.75531E-05	-5.39597E-05	5.98162E-06	-5.36240E-05	1.00032E-06	-5.36240E-05
5.702500	-1.07989E-03	-1.48067E-03	-1.37268E-04	1.75531E-05	-5.39597E-05	5.98162E-06	-5.36240E-05	1.00032E-06	-5.36240E-05	5.802500	-1.06617E-03	-1.50665E-03	-1.38137E-04	1.81285E-05	-5.6906E-05	4.79366E-06	-5.65791E-05	9.99899E-07	-5.65791E-05
5.802500	-1.06617E-03	-1.50665E-03	-1.38137E-04	1.81285E-05	-5.6906E-05	4.79366E-06	-5.65791E-05	9.99899E-07	-5.65791E-05	5.902500	-1.04617E-03	-1.48067E-03	-1.37268E-04	1.75531E-05	-5.39597E-05	5.98162E-06	-5.36240E-05	9.99899E-07	-5.36240E-05
5.902500	-1.09188E-03	-1.53264E-03	-1.39274E-04	1.85826E-05	-5.93156E-05	3.13912E-06	-5.95968E-05	9.99337E-07	-5.95968E-05	6.002500	-1.09188E-03	-1.53264E-03	-1.39274E-04	1.85826E-05	-5.93156E-05	3.13912E-06	-5.95968E-05	9.99337E-07	-5.95968E-05
6.002500	-1.09188E-03	-1.53264E-03	-1.39274E-04	1.85826E-05	-5.93156E-05	3.13912E-06	-5.95968E-05	9.99337E-07	-5.95968E-05	6.102500	-1.09188E-03	-1.53264E-03	-1.39274E-04	1.85826E-05	-5.93156E-05	3.13912E-06	-5.95968E-05	9.99337E-07	-5.95968E-05

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	TIME	THE TAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETA E	PHIE
ANGLE RATES	6.102500	-1.10154E-03 -4.24078E-05	-1.58460E-03 -2.59823E-04	-1.42426E-04 -1.87970E-05	1.94221E-05 3.95816E-06	-6.61196E-05 -3.16245E-05	-1.71719E-06 -2.97085E-05	-6.58139E-05 -3.16800E-05	9.97730E-07 -9.72877E-09
ANGLE RATES	6.202500	-1.10549E-03 -3.66133E-05	-1.61058E-03 -2.59823E-04	-1.44467E-04 -2.19694E-05	1.98052E-05 3.70437E-06	-6.93115E-05 -3.22154E-05	-4.98018E-06 -3.55179E-05	-6.90108E-05 -3.22549E-05	9.96666E-07 -1.15349E-08
ANGLE RATES	6.302500	-1.10886E-03 -3.07500E-05	-1.63657E-03 -2.59823E-04	-1.46827E-04 -2.51690E-05	2.01624E-05 3.43981E-06	-7.25621E-05 -3.27984E-05	-8.83663E-06 -4.15758E-05	-7.22652E-05 -3.28314E-05	9.95418E-07 -1.34119E-08
ANGLE RATES	6.402500	-1.11164E-03 -2.48832E-05	-1.66255E-03 -2.59822E-04	-1.49505E-04 -2.83431E-05	2.04926E-05 3.16474E-06	-7.58707E-05 -3.33755E-05	-1.33095E-05 -4.78456E-05	-7.55765E-05 -3.33966E-05	9.93980E-07 -1.53468E-08
ANGLE RATES	6.502500	-1.11383E-03 -1.90186E-05	-1.68853E-03 -2.59822E-04	-1.52497E-04 -3.14394E-05	2.07948E-05 2.87933E-06	-7.92363E-05 -3.39487E-05	-1.84182E-05 -5.42902E-05	-7.89441E-05 -3.39579E-05	9.92346E-07 -1.73260E-08
ANGLE RATES	6.602500	-1.11544E-03 -1.31616E-05	-1.71451E-03 -2.59822E-04	-1.55792E-04 -3.44066E-05	2.10580E-05 2.58373E-06	-8.26601E-05 -3.45200E-05	-2.41782E-05 -6.08728E-05	-8.23678E-05 -3.45174E-05	9.90512E-07 -1.93356E-08
ANGLE RATES	6.702500	-1.11647E-03 -7.31759E-06	-1.74049E-03 -2.59822E-04	-1.59374E-04 -3.71956E-05	2.13111E-05 2.27805E-06	-8.61406E-05 -3.50911E-05	-3.06015E-05 -6.75569E-05	-8.58474E-05 -3.50774E-05	9.88476E-07 -2.13617E-08
ANGLE RATES	6.802500	-1.11691E-03 -1.49176E-06	-1.76648E-03 -2.59822E-04	-1.63224E-04 -3.97598E-05	2.15231E-05 1.96239E-06	-8.95763E-05 -3.56639E-05	-3.76964E-05 -7.43071E-05	-8.93832E-05 -3.56401E-05	9.86238E-07 -2.33901E-08
ANGLE RATES	6.902500	-1.11677E-03 4.31075E-06	-1.79246E-03 -2.59822E-04	-1.67317E-04 -4.20562E-05	2.17030E-05 1.63638E-06	-9.32734E-05 -3.62399E-05	-4.54680E-05 -8.10896E-05	-9.29755E-05 -3.62074E-05	9.83797E-07 -2.54070E-08
ANGLE RATES	7.002500	-1.11605E-03 1.00850E-05	-1.81844E-03 -2.59822E-04	-1.71624E-04 -4.43457E-05	2.18500E-05 1.30163E-06	-9.69263E-05 -3.68203E-05	-5.39177E-05 -8.78720E-05	-9.66248E-05 -3.67808E-05	9.81156E-07 -2.73985E-08
ANGLE RATES	7.102500	-1.11475E-03 1.58263E-05	-1.84442E-03 -2.59822E-04	-1.76113E-04 -4.56942E-05	2.19629E-05 9.56841E-07	-1.00638E-04 -3.74064E-05	-6.30440E-05 -9.46244E-05	-1.00332E-04 -3.73618E-05	9.78318E-07 -2.93509E-08
ANGLE RATES	7.202500	-1.11288E-03 2.15301E-05	-1.87041E-03 -2.59822E-04	-1.80748E-04 -4.69726E-05	2.20408E-05 6.02711E-07	-1.04408E-04 -3.79988E-05	-7.28425E-05 -1.01319E-04	-1.04097E-04 -3.79512E-05	9.75287E-07 -3.12510E-08
ANGLE RATES	7.302500	-1.11045E-03 2.71922E-05	-1.89639E-03 -2.59822E-04	-1.85491E-04 -4.74578E-05	2.20829E-05 2.39514E-07	-1.08238E-04 -3.85982E-05	-8.33062E-05 -1.07929E-04	-1.07922E-04 -3.85497E-05	9.72070E-07 -3.30856E-08
ANGLE RATES	7.402500	-1.10745E-03 3.28086E-05	-1.92237E-03 -2.59822E-04	-1.90301E-04 -4.83324E-05	2.20893E-05 -1.32429E-07	-1.12128E-04 -3.92047E-05	-9.44254E-05 -1.14432E-04	-1.11808E-04 -3.91573E-05	9.68673E-07 -3.48424E-08
ANGLE RATES	7.502500	-1.10339E-03 3.83755E-05	-1.94835E-03 -2.59822E-04	-1.95137E-04 -4.83855E-05	2.20550E-05 -5.12744E-07	-1.16079E-04 -3.98183E-05	-1.04188E-04 -1.20808E-04	-1.15754E-04 -3.77738E-05	9.65105E-07 -3.65091E-08
ANGLE	7.602500	-1.09977E-03	-1.97433E-03	-1.99953E-04	2.19853E-05	-1.20092E-04	-1.18502E-04	-1.19763E-04	9.61175E-07

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	TIME	THETAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETA	PHIE
ANGLE	7.702500	-1.09511E-03	-2.00032E-03	-2.04719E-04	2.18754E-05	-1.24167E-04	-1.31590E-04	-1.23835E-04	9.57494E-07
RATES		4.93478E-05	-2.59822E-04	-4.72169E-05	-1.29675E-06	-4.10644E-05	-1.33106E-04	-4.10312E-05	-3.95275E-08
ANGLE	7.802500	-1.08991E-03	-2.02630E-03	-2.09380E-04	2.17255E-05	-1.28305E-04	-1.45195E-04	-1.27970E-04	9.53475E-07
RATES		5.47471E-05	-2.59822E-04	-4.60059E-05	-1.69947E-06	-4.16950E-05	-1.38999E-04	-4.16699E-05	-4.08588E-08
ANGLE	7.902500	-1.08415E-03	-2.05228E-03	-2.13899E-04	2.15351E-05	-1.32506E-04	-1.59381E-04	-1.32169E-04	9.49328E-07
RATES		6.00851E-05	-2.59822E-04	-4.43952E-05	-2.10875E-06	-4.23289E-05	-1.44707E-04	-4.23132E-05	-4.20595E-08
ANGLE	8.002500	-1.07789E-03	-2.07826E-03	-2.18238E-04	2.13035E-05	-1.36771E-04	-1.74128E-04	-1.36433E-04	9.45069E-07
RATES		6.53595E-05	-2.59822E-04	-4.24060E-05	-2.52376E-06	-4.29644E-05	-1.50221E-04	-4.29595E-05	-4.31221E-08
ANGLE	8.102500	-1.07110E-03	-2.10425E-03	-2.22360E-04	2.10300E-05	-1.41099E-04	-1.89416E-04	-1.40761E-04	9.40710E-07
RATES		7.05682E-05	-2.59822E-04	-4.00654E-05	-2.94428E-06	-4.35993E-05	-1.55536E-04	-4.36066E-05	-4.40405E-08
ANGLE	8.202500	-1.06378E-03	-2.13023E-03	-2.26232E-04	2.07143E-05	-1.45491E-04	-2.05226E-04	-1.45154E-04	9.36268E-07
RATES		7.57025E-05	-2.59822E-04	-3.74058E-05	-3.36970E-06	-4.42313E-05	-1.60646E-04	-4.42523E-05	-4.48102E-08
ANGLE	8.302500	-1.05595E-03	-2.15621E-03	-2.29824E-04	2.03558E-05	-1.49946E-04	-2.21536E-04	-1.47612E-04	9.31756E-07
RATES		8.07819E-05	-2.59822E-04	-3.44647E-05	-3.79958E-06	-4.48579E-05	-1.65551E-04	-4.48938E-05	-4.54286E-08
ANGLE	8.402500	-1.04763E-03	-2.18219E-03	-2.33109E-04	1.99542E-05	-1.54462E-04	-2.38326E-04	-1.54133E-04	9.27189E-07
RATES		8.57841E-05	-2.59822E-04	-3.12838E-05	-4.23349E-06	-4.54764E-05	-1.70250E-04	-4.55284E-05	-4.58953E-08
ANGLE	8.502500	-1.03880E-03	-2.20817E-03	-2.36066E-04	1.95089E-05	-1.59041E-04	-2.55575E-04	-1.58717E-04	9.22584E-07
RATES		9.07150E-05	-2.59822E-04	-2.79082E-05	-4.67110E-06	-4.60836E-05	-1.74744E-04	-4.61530E-05	-4.62116E-08
ANGLE	8.602500	-1.02949E-03	-2.23416E-03	-2.38679E-04	1.90197E-05	-1.63679E-04	-2.73264E-04	-1.63363E-04	9.17955E-07
RATES		9.55736E-05	-2.59822E-04	-2.43863E-05	-5.11213E-06	-4.66767E-05	-1.79037E-04	-4.67642E-05	-4.63824E-08
ANGLE	8.702500	-1.01969E-03	-2.26014E-03	-2.40934E-04	1.84863E-05	-1.68375E-04	-2.91373E-04	-1.68069E-04	9.13315E-07
RATES		1.00359E-04	-2.59822E-04	-2.07686E-05	-5.55641E-06	-4.72522E-05	-1.83133E-04	-4.73585E-05	-4.64140E-08
ANGLE	8.802500	-1.00942E-03	-2.28612E-03	-2.42825E-04	1.79083E-05	-1.73128E-04	-3.09881E-04	-1.72834E-04	9.08679E-07
RATES		1.05071E-04	-2.59822E-04	-1.71068E-05	-6.00383E-06	-4.78071E-05	-1.87037E-04	-4.79325E-05	-4.63162E-08
ANGLE	8.902500	-9.98682E-04	-2.31210E-03	-2.44351E-04	1.72854E-05	-1.77936E-04	-3.28770E-04	-1.77655E-04	9.04059E-07
RATES		1.09708E-04	-2.59822E-04	-1.34533E-05	-6.45440E-06	-4.83380E-05	-1.90756E-04	-4.84823E-05	-4.61016E-08
ANGLE	9.002500	-9.87483E-04	-2.33809E-03	-2.45514E-04	1.66172E-05	-1.82795E-04	-3.48023E-04	-1.82530E-04	8.99465E-07
RATES		1.14270E-04	-2.59822E-04	-9.86032E-06	-6.90819E-06	-4.88416E-05	-1.94298E-04	-4.90045E-05	-4.57862E-08
ANGLE	9.102500	-9.75831E-04	-2.36407E-03	-2.46323E-04	1.59036E-05	-1.87703E-04	-3.67620E-04	-1.87455E-04	8.94907E-07
RATES		1.18757E-04	-2.59823E-04	-6.37901E-06	-7.36538E-06	-4.93148E-05	-1.97669E-04	-4.94955E-05	-4.53886E-08
ANGLE	9.202500	-9.64273E-04	-2.39005E-03	-2.47132E-04	1.51643E-05	-1.92457E-04	-3.87647E-04	-1.87647E-04	8.90465E-07
RATES		1.23333E-04	-2.59822E-04	-5.10000E-06	-8.00000E-06	-5.00000E-05	-2.00000E-04	-5.00000E-05	-4.50000E-08

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	TIME	THETAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETA	PHIE
ANGLE RATES	7.702500	-1.09511E-03 4.93478E-05	-2.00032E-03 -2.59822E-04	-2.04719E-04 -4.72169E-05	2.18754E-05 -1.29675E-06	-1.24167E-04 -4.10644E-05	-1.31590E-04 -1.33106E-04	-1.23835E-04 -4.10312E-05	9.57494E-07 -3.95275E-08
ANGLE RATES	7.802500	-1.08991E-03 5.47471E-05	-2.02630E-03 -2.59822E-04	-2.09380E-04 -4.60059E-05	2.17255E-05 -1.69947E-06	-1.28305E-04 -4.16950E-05	-1.45195E-04 -1.38999E-04	-1.27970E-04 -4.16699E-05	9.53475E-07 -4.08588E-08
ANGLE RATES	7.902500	-1.08415E-03 6.00891E-05	-2.05228E-03 -2.59822E-04	-2.13899E-04 -4.43952E-05	2.15351E-05 -2.10875E-06	-1.32506E-04 -4.23289E-05	-1.59381E-04 -1.44707E-04	-1.32169E-04 -4.23132E-05	9.49328E-07 -4.20595E-08
ANGLE RATES	8.002500	-1.07749E-03 6.53595E-05	-2.07826E-03 -2.59822E-04	-2.18238E-04 -4.24060E-05	2.13035E-05 -2.52376E-06	-1.36771E-04 -4.29644E-05	-1.74128E-04 -1.50221E-04	-1.35433E-04 -4.29595E-05	9.45069E-07 -4.31221E-08
ANGLE RATES	8.102500	-1.07110E-03 7.05682E-05	-2.10425E-03 -2.59822E-04	-2.22360E-04 -4.00654E-05	2.10300E-05 -2.94428E-06	-1.41099E-04 -4.35993E-05	-1.89416E-04 -1.55536E-04	-1.43761E-04 -4.36066E-05	9.40710E-07 -4.40405E-08
ANGLE RATES	8.202500	-1.06378E-03 7.57075E-05	-2.13023E-03 -2.59822E-04	-2.26232E-04 -3.74058E-05	2.07143E-05 -3.36970E-06	-1.45491E-04 -4.42313E-05	-2.05226E-04 -1.60046E-04	-1.45154E-04 -4.42523E-05	9.36268E-07 -4.48102E-08
ANGLE RATES	8.302500	-1.05596E-03 8.07819E-05	-2.15621E-03 -2.59822E-04	-2.29824E-04 -3.44647E-05	2.03558E-05 -3.79958E-06	-1.49946E-04 -4.43579E-05	-2.21536E-04 -1.65551E-04	-1.43612E-04 -4.48938E-05	9.31756E-07 -4.54286E-08
ANGLE RATES	8.402500	-1.04763E-03 8.57841E-05	-2.18219E-03 -2.59822E-04	-2.33109E-04 -3.12838E-05	1.99542E-05 -4.23349E-06	-1.54462E-04 -4.54764E-05	-2.38326E-04 -1.70250E-04	-1.54133E-04 -4.55284E-05	9.27189E-07 -4.58953E-08
ANGLE RATES	8.502500	-1.03880E-03 9.07150E-05	-2.20817E-03 -2.59822E-04	-2.36066E-04 -2.79082E-05	1.95089E-05 -4.67110E-06	-1.59041E-04 -4.60836E-05	-2.55575E-04 -1.74744E-04	-1.58717E-04 -4.61530E-05	9.22584E-07 -4.62116E-08
ANGLE RATES	8.602500	-1.02949E-03 9.55736E-05	-2.23416E-03 -2.59822E-04	-2.38679E-04 -2.43863E-05	1.90197E-05 -5.11213E-06	-1.63679E-04 -4.66767E-05	-2.73264E-04 -1.79037E-04	-1.63363E-04 -4.67642E-05	9.17955E-07 -4.63824E-08
ANGLE RATES	8.702500	-1.01969E-03 1.00359E-04	-2.26014E-03 -2.59822E-04	-2.40934E-04 -2.07686E-05	1.84863E-05 -5.55641E-06	-1.68375E-04 -4.72522E-05	-2.91373E-04 -1.83133E-04	-1.68069E-04 -4.73585E-05	9.13315E-07 -4.64140E-08
ANGLE RATES	8.802500	-1.00942E-03 1.05071E-04	-2.28612E-03 -2.59822E-04	-2.42825E-04 -1.71068E-05	1.79083E-05 -6.00383E-06	-1.73128E-04 -4.78071E-05	-3.09881E-04 -1.87037E-04	-1.72834E-04 -4.79325E-05	9.08679E-07 -4.63162E-08
ANGLE RATES	8.902500	-9.98682E-04 1.09708E-04	-2.31210E-03 -2.59822E-04	-2.44351E-04 -1.34533E-05	1.72854E-05 -6.45440E-06	-1.77936E-04 -4.83380E-05	-3.28770E-04 -1.90756E-04	-1.77655E-04 -4.84823E-05	9.04059E-07 -4.61016E-08
ANGLE RATES	9.002500	-9.87483E-04 1.14270E-04	-2.33809E-03 -2.59822E-04	-2.45514E-04 -9.86032E-06	1.66172E-05 -6.90819E-06	-1.82795E-04 -4.88416E-05	-3.48023E-04 -1.94298E-04	-1.82530E-04 -4.93045E-05	8.99465E-07 -4.57862E-08
ANGLE RATES	9.102500	-9.75831E-04 1.18757E-04	-2.36407E-03 -2.59823E-04	-2.46323E-04 -6.37901E-06	1.59036E-05 -7.36538E-06	-1.87703E-04 -4.93148E-05	-3.67620E-04 -1.97669E-04	-1.87455E-04 -4.94955E-05	8.94907E-07 -4.53886E-08
ANGLE RATES	9.202500	-9.64235E-04	-2.39004E-03	-2.47102E-04	1.51441E-05	-1.92457E-04	-3.87647E-04	-1.91617E-04	8.90411E-07

	TIME	THETAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETA E	PHIE
ANGLE	10.902500	-6.97898E-04	-2.83175E-03	-2.32433E-04	-5.33244E-06	-2.79680E-04	-7.54389E-04	-2.79830E-04	8.11893E-07
RATES		1.85291E-04	-2.59825E-04	2.29017E-06	-1.66428E-05	-5.03165E-05	-2.37953E-04	-5.05367E-05	-6.05159E-08
ANGLE	11.002500	-6.79230E-04	-2.85773E-03	-2.32337E-04	-7.02676E-06	-2.84692E-04	-7.88257E-04	-2.84913E-04	8.05622E-07
RATES		1.88073E-04	-2.59825E-04	-3.29137E-07	-1.72439E-05	-4.99279E-05	-2.39402E-04	-5.01342E-05	-6.49299E-08
ANGLE	11.102500	-6.60289E-04	-2.88372E-03	-2.32510E-04	-8.78176E-06	-2.89664E-04	-8.12266E-04	-2.89904E-04	7.98880E-07
RATES		1.90740E-04	-2.59825E-04	-3.08837E-06	-1.78561E-05	-4.95007E-05	-2.40783E-04	-4.96923E-05	-6.99292E-08
ANGLE	11.202500	-6.41087E-04	-2.90970E-03	-2.32964E-04	-1.05995E-05	-2.94591E-04	-8.36410E-04	-2.94849E-04	7.91608E-07
RATES		1.93289E-04	-2.59825E-04	-5.93747E-06	-1.84796E-05	-4.90374E-05	-2.42096E-04	-4.92138E-05	-7.55297E-08
ANGLE	11.302500	-6.21637E-04	-2.93568E-03	-2.33705E-04	-1.24782E-05	-2.99469E-04	-8.60682E-04	-2.99745E-04	7.83745E-07
RATES		1.95716E-04	-2.59825E-04	-8.82494E-06	-1.91144E-05	-4.85404E-05	-2.43340E-04	-4.87016E-05	-8.17433E-08
ANGLE	11.402500	-6.01950E-04	-2.96166E-03	-2.34733E-04	-1.44220E-05	-3.04297E-04	-8.85075E-04	-3.04588E-04	7.75229E-07
RATES		1.98018E-04	-2.59825E-04	-1.16986E-05	-1.97606E-05	-4.80125E-05	-2.44513E-04	-4.81586E-05	-8.85782E-08
ANGLE	11.502500	-5.82039E-04	-2.98765E-03	-2.36046E-04	-1.64309E-05	-3.09070E-04	-9.09581E-04	-3.09375E-04	7.65999E-07
RATES		2.00193E-04	-2.59825E-04	-1.45063E-05	-2.04180E-05	-4.74564E-05	-2.45613E-04	-4.75876E-05	-9.60389E-08
ANGLE	11.602500	-5.61918E-04	-3.01363E-03	-2.37634E-04	-1.85062E-05	-3.13787E-04	-9.34194E-04	-3.14104E-04	7.55990E-07
RATES		2.02236E-04	-2.59826E-04	-1.71972E-05	-2.10866E-05	-4.68745E-05	-2.46639E-04	-4.69914E-05	-1.04126E-07
ANGLE	11.702500	-5.41598E-04	-3.03961E-03	-2.39482E-04	-2.06488E-05	-3.18444E-04	-9.58906E-04	-3.18772E-04	7.45142E-07
RATES		2.04146E-04	-2.59826E-04	-1.97217E-05	-2.17661E-05	-4.62695E-05	-2.47588E-04	-4.63729E-05	-1.12839E-07
ANGLE	11.802500	-5.21095E-04	-3.06559E-03	-2.41572E-04	-2.28599E-05	-3.23039E-04	-9.83708E-04	-3.23377E-04	7.33391E-07
RATES		2.05919E-04	-2.59826E-04	-2.20331E-05	-2.24563E-05	-4.56438E-05	-2.48459E-04	-4.57345E-05	-1.22173E-07
ANGLE	11.902500	-5.00421E-04	-3.09158E-03	-2.43881E-04	-2.51406E-05	-3.27571E-04	-1.00859E-03	-3.27918E-04	7.20676E-07
RATES		2.07554E-04	-2.59826E-04	-2.40879E-05	-2.31567E-05	-4.49996E-05	-2.49248E-04	-4.50787E-05	-1.32124E-07
ANGLE	12.002500	-4.79591E-04	-3.11756E-03	-2.46379E-04	-2.74918E-05	-3.32038E-04	-1.03355E-03	-3.32392E-04	7.06935E-07
RATES		2.09047E-04	-2.59826E-04	-2.58465E-05	-2.38670E-05	-4.43391E-05	-2.49955E-04	-4.44077E-05	-1.42683E-07
ANGLE	12.102500	-4.58619E-04	-3.14354E-03	-2.49037E-04	-2.99145E-05	-3.36438E-04	-1.05858E-03	-3.36798E-04	6.92109E-07
RATES		2.10397E-04	-2.59826E-04	-2.72738E-05	-2.45866E-05	-4.36641E-05	-2.50575E-04	-4.47235E-05	-1.53847E-07
ANGLE	12.202500	-4.37519E-04	-3.16952E-03	-2.51819E-04	-3.24096E-05	-3.40770E-04	-1.08366E-03	-3.41136E-04	6.76136E-07
RATES		2.11602E-04	-2.59826E-04	-2.83399E-05	-2.53149E-05	-4.29762E-05	-2.51109E-04	-4.30278E-05	-1.65608E-07
ANGLE	12.302500	-4.16306E-04	-3.19551E-03	-2.54589E-04	-3.49780E-05	-3.45033E-04	-1.10880E-03	-3.45403E-04	6.58957E-07
RATES		2.12661E-04	-2.59827E-04	-2.90205E-05	-2.60514E-05	-4.22769E-05	-2.51554E-04	-4.23222E-05	-1.77964E-07
ANGLE	12.402500	-3.94996E-04	-3.22149E-03	-2.57505E-04	-3.76773E-05	-3.49776E-04	-1.13367E-03	-3.49600E-04	6.40513E-07

	TIME	THETAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETAE	PHIE
ANGLE	12.502500	-3.73598E-04	-3.24747E-03	-2.60529E-04	-4.03374E-05	-3.53346E-04	-1.15917E-03	-3.53724E-04	6.20744E-07
RATES		2.14336E-04	-2.59827E-04	-2.91576E-05	-2.75461E-05	-4.08478E-05	-2.52169E-04	-4.08855E-05	-2.04462E-07
ANGLE	12.602500	-3.52134E-04	-3.27345E-03	-2.63417E-04	-4.31299E-05	-3.57394E-04	-1.18440E-03	-3.57776E-04	5.99591E-07
RATES		2.14951E-04	-2.59827E-04	-2.85962E-05	-2.83032E-05	-4.01192E-05	-2.52339E-04	-4.01560E-05	-2.18614E-07
ANGLE	12.702500	-3.30616E-04	-3.29944E-03	-2.66227E-04	-4.59934E-05	-3.61369E-04	-1.20964E-03	-3.61755E-04	5.76991E-07
RATES		2.15417E-04	-2.59827E-04	-2.76138E-05	-2.90659E-05	-3.93815E-05	-2.52414E-04	-3.94192E-05	-2.33385E-07
ANGLE	12.802500	-3.09058E-04	-3.32542E-03	-2.68918E-04	-4.89434E-05	-3.65270E-04	-1.23498E-03	-3.65660E-04	5.52883E-07
RATES		2.15736E-04	-2.59827E-04	-2.62177E-05	-2.98338E-05	-3.86346E-05	-2.52396E-04	-3.86752E-05	-2.48794E-07
ANGLE	12.902500	-2.87476E-04	-3.35140E-03	-2.71450E-04	-5.17654E-05	-3.69095E-04	-1.26011E-03	-3.69490E-04	5.27202E-07
RATES		2.15907E-04	-2.59827E-04	-2.44215E-05	-3.06064E-05	-3.78779E-05	-2.52285E-04	-3.79232E-05	-2.64868E-07
ANGLE	13.002500	-2.65884E-04	-3.37738E-03	-2.73782E-04	-5.50650E-05	-3.72845E-04	-1.28533E-03	-3.73244E-04	4.99878E-07
RATES		2.15932E-04	-2.59827E-04	-2.22452E-05	-3.13837E-05	-3.71105E-05	-2.52080E-04	-3.71625E-05	-2.81640E-07
ANGLE	13.102500	-2.44297E-04	-3.40337E-03	-2.75879E-04	-5.82424E-05	-3.76517E-04	-1.31052E-03	-3.74922E-04	4.70841E-07
RATES		2.15811E-04	-2.59827E-04	-1.97145E-05	-3.21653E-05	-3.63315E-05	-2.51784E-04	-3.63916E-05	-2.99153E-07
ANGLE	13.202500	-2.22729E-04	-3.42935E-03	-2.77706E-04	-6.14982E-05	-3.80111E-04	-1.33568E-03	-3.80522E-04	4.40313E-07
RATES		2.15547E-04	-2.59827E-04	-1.68606E-05	-3.29514E-05	-3.55394E-05	-2.51396E-04	-3.56091E-05	-3.17456E-07
ANGLE	13.302500	-2.01195E-04	-3.45533E-03	-2.79233E-04	-6.48329E-05	-3.83624E-04	-1.36080E-03	-3.84043E-04	4.07312E-07
RATES		2.15141E-04	-2.59827E-04	-1.37197E-05	-3.37421E-05	-3.47326E-05	-2.50920E-04	-3.48132E-05	-3.36610E-07
ANGLE	13.402500	-1.79708E-04	-3.48132E-03	-2.80434E-04	-6.82469E-05	-3.87056E-04	-1.38586E-03	-3.87484E-04	3.72651E-07
RATES		2.14596E-04	-2.59828E-04	-1.03326E-05	-3.45380E-05	-3.39094E-05	-2.50356E-04	-3.40017E-05	-3.56680E-07
ANGLE	13.502500	-1.58283E-04	-3.50730E-03	-2.81285E-04	-7.17407E-05	-3.90405E-04	-1.41087E-03	-3.90843E-04	3.35934E-07
RATES		2.13914E-04	-2.59828E-04	-6.74351E-06	-3.53394E-05	-3.30677E-05	-2.49707E-04	-3.31724E-05	-3.77740E-07
ANGLE	13.602500	-1.36932E-04	-3.53328E-03	-2.81770E-04	-7.53150E-05	-3.93669E-04	-1.43580E-03	-3.94118E-04	2.97057E-07
RATES		2.13097E-04	-2.59828E-04	-3.06002E-06	-3.61471E-05	-3.22054E-05	-2.48975E-04	-3.23230E-05	-3.99874E-07
ANGLE	13.702500	-1.15670E-04	-3.55926E-03	-2.81875E-04	-7.89704E-05	-3.96846E-04	-1.46066E-03	-3.97307E-04	2.55910E-07
RATES		2.12149E-04	-2.59828E-04	8.48019E-07	-3.69619E-05	-3.13202E-05	-2.48164E-04	-3.4511E-05	-4.23169E-07
ANGLE	13.802500	-9.45092E-05	-3.58525E-03	-2.81593E-04	-8.27077E-05	-3.99932E-04	-1.48543E-03	-4.00407E-04	2.12370E-07
RATES		2.11072E-04	-2.59828E-04	4.74919E-06	-3.77848E-05	-3.04099E-05	-2.47277E-04	-3.05540E-05	-4.47719E-07
ANGLE	13.902500	-7.34623E-05	-3.61123E-03	-2.80920E-04	-8.65278E-05	-4.02926E-04	-1.51011E-03	-4.03416E-04	1.66308E-07
RATES		2.09869E-04	-2.59828E-04	8.65129E-06	-3.86168E-05	-2.94721E-05	-2.46315E-04	-2.96293E-05	-4.73625E-07
ANGLE	14.002500	-5.25419E-05	-3.63721E-03	-2.79860E-04	-9.04315E-05	-4.05625E-04	-1.53469E-03	-4.05332E-04	1.17582E-07

	TIME	THETAB	PHIB	THETAS	PHIS	THETAP	PIIP	THETAE	PHIE
ANGLE	12.502500	-3.73598E-04	-3.24747E-03	-2.60529E-04	-4.03374E-05	-3.53346E-04	-1.15917E-03	-3.53724E-04	6.20744E-07
RATES		2.14336E-04	-2.59827E-04	-2.91576E-05	-2.75461E-05	-4.08478E-05	-2.52169E-04	-4.08855E-05	-2.04462E-07
ANGLE	12.602500	-3.52134E-04	-3.27345E-03	-2.63417E-04	-4.31299E-05	-3.57394E-04	-1.18440E-03	-3.57776E-04	5.99591E-07
RATES		2.14951E-04	-2.59827E-04	-2.85962E-05	-2.83032E-05	-4.01192E-05	-2.52339E-04	-4.01560E-05	-2.18614E-07
ANGLE	12.702500	-3.30616E-04	-3.29944E-03	-2.66227E-04	-4.59934E-05	-3.61309E-04	-1.20964E-03	-3.61755E-04	5.76991E-07
RATES		2.15417E-04	-2.59827E-04	-2.76138E-05	-2.90659E-05	-3.93815E-05	-2.52414E-04	-3.94192E-05	-2.33385E-07
ANGLE	12.802500	-3.09058E-04	-3.32542E-03	-2.68918E-04	-4.89434E-05	-3.65270E-04	-1.23498E-03	-3.65660E-04	5.52883E-07
RATES		2.15736E-04	-2.59827E-04	-2.62177E-05	-2.98338E-05	-3.86346E-05	-2.52396E-04	-3.86752E-05	-2.48794E-07
ANGLE	12.902500	-2.87476E-04	-3.35140E-03	-2.71450E-04	-5.17654E-05	-3.69095E-04	-1.26011E-03	-3.69490E-04	5.27202E-07
RATES		2.15907E-04	-2.59827E-04	-2.44215E-05	-3.06064E-05	-3.78779E-05	-2.52205E-04	-3.79232E-05	-2.64868E-07
ANGLE	13.002500	-2.65884E-04	-3.37738E-03	-2.73782E-04	-5.50650E-05	-3.72845E-04	-1.28533E-03	-3.73244E-04	4.99878E-07
RATES		2.15932E-04	-2.59827E-04	-2.22452E-05	-3.13837E-05	-3.71105E-05	-2.52080E-04	-3.71625E-05	-2.81640E-07
ANGLE	13.102500	-2.44297E-04	-3.40337E-03	-2.75879E-04	-5.82424E-05	-3.76517E-04	-1.31052E-03	-3.74922E-04	4.70841E-07
RATES		2.15811E-04	-2.59827E-04	-1.97145E-05	-3.21633E-05	-3.63315E-05	-2.51784E-04	-3.63916E-05	-2.99153E-07
ANGLE	13.202500	-2.22729E-04	-3.42935E-03	-2.77706E-04	-6.14982E-05	-3.80111E-04	-1.33568E-03	-3.80522E-04	4.40313E-07
RATES		2.15547E-04	-2.59827E-04	-1.68606E-05	-3.29514E-05	-3.55394E-05	-2.51396E-04	-3.56091E-05	-3.17456E-07
ANGLE	13.302500	-2.01195E-04	-3.45533E-03	-2.79233E-04	-6.48329E-05	-3.83624E-04	-1.36080E-03	-3.84043E-04	4.07312E-07
RATES		2.15141E-04	-2.59827E-04	-1.37197E-05	-3.37421E-05	-3.47326E-05	-2.50920E-04	-3.48132E-05	-3.36610E-07
ANGLE	13.402500	-1.79708E-04	-3.48132E-03	-2.80434E-04	-6.82469E-05	-3.87056E-04	-1.38586E-03	-3.87484E-04	3.72651E-07
RATES		2.14596E-04	-2.59828E-04	-1.03326E-05	-3.45380E-05	-3.39094E-05	-2.50356E-04	-3.40017E-05	-3.56680E-07
ANGLE	13.502500	-1.58283E-04	-3.50730E-03	-2.81285E-04	-7.17407E-05	-3.90405E-04	-1.41087E-03	-3.90843E-04	3.35934E-07
RATES		2.13914E-04	-2.59828E-04	-6.74351E-06	-3.53394E-05	-3.30677E-05	-2.49707E-04	-3.31724E-05	-3.77740E-07
ANGLE	13.602500	-1.36932E-04	-3.53328E-03	-2.81770E-04	-7.53150E-05	-3.93669E-04	-1.43580E-03	-3.94118E-04	2.97057E-07
RATES		2.13097E-04	-2.59828E-04	-3.00002E-06	-3.61471E-05	-3.22054E-05	-2.48975E-04	-3.23230E-05	-3.99874E-07
ANGLE	13.702500	-1.15670E-04	-3.55926E-03	-2.81875E-04	-7.89704E-05	-3.96846E-04	-1.46066E-03	-3.97307E-04	2.55910E-07
RATES		2.12149E-04	-2.59828E-04	8.48019E-07	-3.69619E-05	-3.13202E-05	-2.48164E-04	-3.4511E-05	-4.23169E-07
ANGLE	13.802500	-9.45092E-05	-3.58525E-03	-2.81593E-04	-8.27077E-05	-3.99932E-04	-1.48543E-03	-4.00407E-04	2.12370E-07
RATES		2.11072E-04	-2.59828E-04	4.74919E-06	-3.77848E-05	-3.04099E-05	-2.47277E-04	-3.05540E-05	-4.47719E-07
ANGLE	13.902500	-7.34623E-05	-3.61123E-03	-2.80920E-04	-8.65278E-05	-4.02926E-04	-1.51011E-03	-4.03416E-04	1.66308E-07
RATES		2.09869E-04	-2.59828E-04	8.65129E-06	-3.86168E-05	-2.94721E-05	-2.46315E-04	-2.96293E-05	-4.73625E-07
ANGLE	14.002500	-5.25419E-05	-3.63721E-03	-2.79860E-04	-9.04315E-05	-4.05625E-04	-1.53469E-03	-4.05332E-04	1.17587E-07

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	TIME	THETAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETA E	PHIE
ANGLE RATES	14.102500	-3.17600E-05 2.07098E-04	-3.66320E-03 -2.59828E-04	-2.78423E-04 1.62505E-05	-9.44200E-05 -4.03122E-05	-4.08626E-04 -2.75050E-05	-1.55916E-03 -2.44185E-04	-4.09150E-04 -2.76871E-05	6.60424E-08 -5.29911E-07
ANGLE RATES	14.202500	-1.11285E-05 2.05536E-04	-3.68918E-03 -2.59828E-04	-2.76612E-04 1.98467E-05	-9.84944E-05 -4.11776E-05	-4.11325E-04 -2.64713E-05	-1.58357E-03 -2.43022E-04	-4.11867E-04 -2.66649E-05	1.15270E-08 -5.60502E-07
ANGLE RATES	14.302500	9.34125E-06 2.03862E-04	-3.71516E-03 -2.59828E-04	-2.74455E-04 2.32434E-05	-1.02656E-04 -4.20560E-05	-4.13919E-04 -2.54015E-05	-1.60776E-03 -2.41799E-04	-4.14481E-04 -2.56057E-05	-4.61360E-08 -5.92861E-07
ANGLE RATES	14.402500	2.96380E-05 2.02077E-04	-3.74114E-03 -2.59828E-04	-2.71971E-04 2.63967E-05	-1.06906E-04 -4.29482E-05	-4.16403E-04 -2.42937E-05	-1.63188E-03 -2.40518E-04	-4.16987E-04 -2.45074E-05	-1.07129E-07 -6.27090E-07
ANGLE RATES	14.502500	4.97509E-05 2.00185E-04	-3.76713E-03 -2.59828E-04	-2.69186E-04 2.92662E-05	-1.11246E-04 -4.38549E-05	-4.18776E-04 -2.31465E-05	-1.65586E-03 -2.39183E-04	-4.19381E-04 -2.33682E-05	-1.71642E-07 -6.63284E-07
ANGLE RATES	14.602500	6.96695E-05 1.98189E-04	-3.79311E-03 -2.59828E-04	-2.66133E-04 3.18163E-05	-1.15678E-04 -4.47765E-05	-4.21031E-04 -2.19584E-05	-1.67971E-03 -2.37796E-04	-4.21659E-04 -2.21867E-05	-2.39879E-07 -7.01531E-07
ANGLE RATES	14.702500	8.93833E-05 1.96091E-04	-3.81909E-03 -2.59828E-04	-2.62837E-04 3.46163E-05	-1.20202E-04 -4.57131E-05	-4.23165E-04 -2.07284E-05	-1.70342E-03 -2.36360E-04	-4.23816E-04 -2.09615E-05	-3.12047E-07 -7.41912E-07
ANGLE RATES	14.802500	1.08882E-04 1.93895E-04	-3.84507E-03 -2.59828E-04	-2.59343E-04 3.8408E-05	-1.24821E-04 -4.66648E-05	-4.25174E-04 -1.94557E-05	-1.72698E-03 -2.34877E-04	-4.25849E-04 -1.96917E-05	-3.88364E-07 -7.94496E-07
ANGLE RATES	14.902500	1.29157E-04 1.91601E-04	-3.87106E-03 -2.59828E-04	-2.56866E-04 3.72706E-05	-1.29536E-04 -4.76314E-05	-4.27054E-04 -1.81400E-05	-1.75039E-03 -2.33349E-04	-4.27752E-04 -1.83769E-05	-4.69053E-07 -8.29340E-07
ANGLE RATES	15.002500	1.47198E-04 1.89214E-04	-3.89704E-03 -2.59828E-04	-2.51907E-04 3.82925E-05	-1.34348E-04 -4.86124E-05	-4.28800E-04 -1.67812E-05	-1.77365E-03 -2.31777E-04	-4.29522E-04 -1.70169E-05	-5.54342E-07 -8.76489E-07
ANGLE RATES	15.102500	1.65995E-04 1.86735E-04	-3.92302E-03 -2.59828E-04	-2.48047E-04 3.88998E-05	-1.39259E-04 -4.96071E-05	-4.30408E-04 -1.53794E-05	-1.79674E-03 -2.30163E-04	-4.31153E-04 -1.56118E-05	-6.44464E-07 -9.25970E-07
ANGLE RATES	15.202500	1.84540E-04 1.84166E-04	-3.94901E-03 -2.59827E-04	-2.44148E-04 3.90922E-05	-1.44270E-04 -5.06146E-05	-4.31874E-04 -1.39355E-05	-1.81968E-03 -2.28508E-04	-4.32642E-04 -1.41624E-05	-7.39651E-07 -9.77798E-07
ANGLE RATES	15.302500	2.02824E-04 1.81508E-04	-3.97499E-03 -2.59827E-04	-2.40250E-04 3.88759E-05	-1.49333E-04 -5.16340E-05	-4.33193E-04 -1.24502E-05	-1.84244E-03 -2.26811E-04	-4.33984E-04 -1.26697E-05	-8.40140E-07 -1.03197E-06
ANGLE RATES	15.402500	2.20837E-04 1.78763E-04	-4.00097E-03 -2.59827E-04	-2.36393E-04 3.82636E-05	-1.54598E-04 -5.26640E-05	-4.34362E-04 -1.09250E-05	-1.86504E-03 -2.25073E-04	-4.35174E-04 -1.11351E-05	-9.46162E-07 -1.08846E-06
ANGLE RATES	15.502500	2.38572E-04 1.75932E-04	-4.02695E-03 -2.59827E-04	-2.32617E-04 3.72741E-05	-1.59916E-04 -5.37033E-05	-4.35376E-04 -9.36159E-06	-1.88746E-03 -2.23293E-04	-4.36208E-04 -9.56049E-06	-1.05795E-06 -1.14722E-06
ANGLE RATES	15.602500	2.56011E-04 1.73101E-04	-4.05293E-03 -2.59827E-04	-2.28901E-04 3.59000E-05	-1.65333E-04 -5.47333E-05	-4.36511E-04 -7.77000E-06	-1.90971E-03 -2.21511E-04	-4.37301E-04 -8.77000E-06	-1.17700E-06 -1.27700E-06

	TIME	THETAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETAE	PHIE
ANGLE	15.702500	2.73171E-04	-4.07892E-03	-2.25450E-04	-1.70867E-04	-4.35927E-04	-1.93175E-03	-4.37796E-04	-1.29970E-06
RATES		1.70018E-04	-2.59827E-04	3.42675E-05	-5.59040E-05	-6.12813E-06	-2.19603E-04	-6.29986E-06	-1.27134E-06
ANGLE	15.802500	2.90019E-04	-4.10490E-03	-2.22123E-04	-1.76500E-04	-4.37456E-04	-1.95361E-03	-4.33342E-04	-1.43010E-06
RATES		1.66936E-04	-2.59827E-04	3.23160E-05	-5.68624E-05	-4.46299E-06	-2.17688E-04	-4.61906E-06	-1.33651E-06
ANGLE	15.902500	3.06554E-04	-4.13088E-03	-2.19033E-04	-1.82240E-04	-4.37818E-04	-1.97528E-03	-4.36718E-04	-1.56711E-06
RATES		1.63773E-04	-2.59827E-04	3.01170E-05	-5.79241E-05	-2.76924E-06	-2.15725E-04	-2.90851E-06	-1.40362E-06
ANGLE	16.002500	3.22769E-04	-4.15687E-03	-2.16114E-04	-1.88085E-04	-4.38008E-04	-1.99676E-03	-4.38922E-04	-1.71093E-06
RATES		1.60527E-04	-2.59827E-04	2.77141E-05	-5.89876E-05	-1.04992E-06	-2.13710E-04	-1.17143E-06	-1.47254E-06
ANGLE	16.102500	3.38655E-04	-4.18285E-03	-2.13472E-04	-1.94037E-04	-4.38026E-04	-2.01802E-03	-4.38951E-04	-1.86172E-06
RATES		1.57201E-04	-2.59826E-04	2.51541E-05	-6.00513E-05	6.91761E-07	-2.11640E-04	5.88807E-07	-1.54312E-06
ANGLE	16.202500	3.54205E-04	-4.20883E-03	-2.11093E-04	-2.00096E-04	-4.37869E-04	-2.03908E-03	-4.38903E-04	-2.01964E-06
RATES		1.53794E-04	-2.59826E-04	2.24859E-05	-6.11137E-05	2.45240E-06	-2.09512E-04	2.36865E-06	-1.61521E-06
ANGLE	16.302500	3.69410E-04	-4.23482E-03	-2.08983E-04	-2.06260E-04	-4.37535E-04	-2.05992E-03	-4.38476E-04	-2.18484E-06
RATES		1.50307E-04	-2.59826E-04	1.97605E-05	-6.21732E-05	4.22873E-06	-2.07320E-04	4.16443E-06	-1.58864E-06
ANGLE	16.402500	3.84262E-04	-4.26080E-03	-2.07146E-04	-2.12530E-04	-4.37622E-04	-2.08054E-03	-4.37969E-04	-2.35745E-06
RATES		1.46740E-04	-2.59826E-04	1.70294E-05	-6.32293E-05	6.01705E-06	-2.09062E-04	5.97244E-06	-1.76323E-06
ANGLE	16.502500	3.98754E-04	-4.28678E-03	-2.05580E-04	-2.18906E-04	-4.36330E-04	-2.10093E-03	-4.37281E-04	-2.53756E-06
RATES		1.43094E-04	-2.59826E-04	1.43444E-05	-6.42776E-05	7.81398E-06	-2.02731E-04	7.73891E-06	-1.83861E-06
ANGLE	16.602500	4.12877E-04	-4.31276E-03	-2.04279E-04	-2.25336E-04	-4.35459E-04	-2.12109E-03	-4.36411E-04	-2.72927E-06
RATES		1.39370E-04	-2.59826E-04	1.17563E-05	-6.53194E-05	9.61595E-06	-2.00324E-04	9.61012E-06	-1.91518E-06
ANGLE	16.702500	4.26624E-04	-4.33875E-03	-2.03227E-04	-2.31969E-04	-4.34407E-04	-2.14099E-03	-4.35358E-04	-2.92064E-06
RATES		1.35568E-04	-2.59825E-04	9.31464E-06	-6.63523E-05	1.14796E-05	-1.97835E-04	1.14324E-05	-1.99216E-06
ANGLE	16.802500	4.39987E-04	-4.36473E-03	-2.02410E-04	-2.38656E-04	-4.33175E-04	-2.16065E-03	-4.34124E-04	-3.12374E-06
RATES		1.31686E-04	-2.59825E-04	7.06635E-06	-6.73747E-05	1.32216E-05	-1.95259E-04	1.32521E-05	-2.06955E-06
ANGLE	16.902500	4.52958E-04	-4.39071E-03	-2.01806E-04	-2.45444E-04	-4.31762E-04	-2.18004E-03	-4.32708E-04	-3.33458E-06
RATES		1.27733E-04	-2.59825E-04	5.05543E-06	-6.83850E-05	1.50188E-05	-1.92569E-04	1.50660E-05	-2.14718E-06
ANGLE	17.002500	4.65530E-04	-4.41669E-03	-2.01389E-04	-2.52332E-04	-4.30171E-04	-2.19916E-03	-4.31111E-04	-3.55319E-06
RATES		1.23703E-04	-2.59825E-04	3.32209E-06	-6.93817E-05	1.68083E-05	-1.89820E-04	1.68709E-05	-2.22486E-06
ANGLE	17.102500	4.77695E-04	-4.44268E-03	-2.01130E-04	-2.59319E-04	-4.28401E-04	-2.21800E-03	-4.29334E-04	-3.77957E-06
RATES		1.19600E-04	-2.59825E-04	1.90215E-06	-7.03632E-05	1.85672E-05	-1.86946E-04	1.86637E-05	-2.30241E-06
ANGLE	17.202500	4.89446E-04	-4.46866E-03	-2.00999E-04	-2.66404E-04	-4.26456E-04	-2.23655E-03	-4.27778E-04	-4.01301E-06

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	TIME	THETAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETA E	PHIE
ANGLE RATES	17.332500	5.00776E-04 1.11180E-04	-4.49464E-03 -2.59824E-04	-2.00949E-04 1.20535E-07	-2.73584E-04 -7.22742E-05	-4.24331E-04 2.21041E-05	-2.25479E-03 -1.80856E-04	-4.25246E-04 2.22032E-05	-4.25549E-06 -2.45649E-06
ANGLE RATES	17.402500	5.11679E-04 1.06868E-04	-4.52062E-03 -2.57824E-04	-2.00953E-04 -1.96112E-07	-2.80958E-04 -7.32055E-05	-4.22034E-04 2.38379E-05	-2.27271E-03 -1.77628E-04	-4.22939E-04 2.37455E-05	-4.50496E-06 -2.53273E-06
ANGLE RATES	17.502500	5.22146E-04 1.02490E-04	-4.54661E-03 -2.59824E-04	-2.00969E-04 -1.09995E-07	-2.88223E-04 -7.41053E-05	-4.19564E-04 2.55530E-05	-2.29031E-03 -1.74268E-04	-4.20458E-04 2.56673E-05	-4.76202E-06 -2.60826E-06
ANGLE RATES	17.502500	5.32173E-04 9.80500E-05	-4.57259E-03 -2.59824E-04	-2.00956E-04 3.85987E-07	-2.95678E-04 -7.49873E-05	-4.15924E-04 2.72481E-05	-2.30756E-03 -1.70769E-04	-4.17806E-04 2.73672E-05	-5.02659E-06 -2.68297E-06
ANGLE RATES	17.702500	5.41753E-04 9.35503E-05	-4.59857E-03 -2.59823E-04	-2.00872E-04 1.29247E-06	-3.03220E-04 -7.58452E-05	-4.14115E-04 2.89224E-05	-2.32445E-03 -1.67124E-04	-4.14985E-04 2.90444E-05	-5.29858E-06 -2.75677E-06
ANGLE RATES	17.802500	5.50880E-04 8.89944E-05	-4.62455E-03 -2.59823E-04	-2.00677E-04 2.00360E-06	-3.10846E-04 -7.66778E-05	-4.11160E-04 3.05753E-05	-2.34098E-03 -1.63326E-04	-4.11998E-04 3.05983E-05	-5.57790E-06 -2.82957E-06
ANGLE RATES	17.902500	5.59549E-04 8.43860E-05	-4.65053E-03 -2.59823E-04	-2.00331E-04 4.30716E-06	-3.18554E-04 -7.74843E-05	-4.08001E-04 3.22064E-05	-2.35711E-03 -1.59367E-04	-4.03847E-04 3.23287E-05	-5.86445E-06 -2.90131E-06
ANGLE RATES	18.002500	5.67755E-04 7.97287E-05	-4.67652E-03 -2.59823E-04	-1.99795E-04 6.38470E-06	-3.26342E-04 -7.82640E-05	-4.04700E-04 3.38158E-05	-2.37284E-03 -1.55238E-04	-4.05534E-04 3.31357E-05	-6.15811E-06 -2.97195E-06
ANGLE RATES	18.102500	5.75492E-04 7.50268E-05	-4.70250E-03 -2.59823E-04	-1.99034E-04 8.81188E-06	-3.34206E-04 -7.90163E-05	-4.01239E-04 3.54039E-05	-2.38815E-03 -1.50931E-04	-4.02061E-04 3.55198E-05	-6.45878E-06 -3.04145E-06
ANGLE RATES	18.202500	5.82757E-04 7.02046E-05	-4.72848E-03 -2.59822E-04	-1.98014E-04 1.15588E-05	-3.42144E-04 -7.97411E-05	-3.97620E-04 3.69714E-05	-2.40302E-03 -1.46437E-04	-3.98431E-04 3.70818E-05	-6.76635E-06 -3.10980E-06
ANGLE RATES	18.302500	5.89547E-04 6.55067E-05	-4.75446E-03 -2.59822E-04	-1.96705E-04 1.45905E-05	-3.50153E-04 -8.04384E-05	-3.93845E-04 3.85192E-05	-2.41743E-03 -1.41746E-04	-3.94646E-04 3.86227E-05	-7.08069E-06 -3.17698E-06
ANGLE RATES	18.402500	5.95857E-04 6.06978E-05	-4.78045E-03 -2.59822E-04	-1.95080E-04 1.78076E-05	-3.58230E-04 -8.11084E-05	-3.89918E-04 4.00483E-05	-2.43136E-03 -1.36848E-04	-3.90707E-04 4.01438E-05	-7.40169E-06 -3.24301E-06
ANGLE RATES	18.502500	6.01685E-04 5.58628E-05	-4.80643E-03 -2.59822E-04	-1.93117E-04 2.13466E-05	-3.66373E-04 -8.17517E-05	-3.85837E-04 4.15603E-05	-2.44479E-03 -1.31731E-04	-3.84618E-04 4.06467E-05	-7.72923E-06 -3.30790E-06
ANGLE RATES	18.602500	6.07029E-04 5.10069E-05	-4.83241E-03 -2.59821E-04	-1.90799E-04 2.49809E-05	-3.74579E-04 -8.23689E-05	-3.81607E-04 4.30565E-05	-2.45770E-03 -1.26384E-04	-3.82379E-04 4.31330E-05	-8.06321E-06 -3.37168E-06
ANGLE RATES	18.702500	6.11885E-04 4.61351E-05	-4.85839E-03 -2.59821E-04	-1.88111E-04 2.87219E-05	-3.82845E-04 -8.29610E-05	-3.77227E-04 4.45388E-05	-2.47006E-03 -1.20793E-04	-3.77992E-04 4.46947E-05	-8.40351E-06 -3.43438E-06
ANGLE RATES	18.802500	6.16254E-04	-4.88437E-03	-1.85167E-04	-3.91170E-04	-3.72700E-04	-2.48184E-03	-3.73600E-04	-8.78000E-06

	TIME	THE TAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETAE	PHIE
ANGLE	18.902500	6.20134E-04	-4.91036E-03	-1.81602E-04	-3.99550E-04	-3.68026E-04	-2.49303E-03	-3.68780E-04	-9.10265E-06
RATES		3.63651E-05	-2.59821E-04	3.63164E-05	-8.40734E-05	4.74681E-05	-1.08820E-04	4.75124E-05	-3.55660E-06
ANGLE	19.002500	6.23525E-04	-4.93634E-03	-1.77780E-04	-4.07983E-04	-3.63207E-04	-2.50359E-03	-3.63957E-04	-9.46129E-06
RATES		3.14774E-05	-2.59821E-04	4.00671E-05	-8.45960E-05	4.89189E-05	-1.02405E-04	4.89524E-05	-3.61622E-06
ANGLE	19.102500	6.26430E-04	-4.96232E-03	-1.73589E-04	-4.16468E-04	-3.58243E-04	-2.51350E-03	-3.58990E-04	-9.82585E-06
RATES		2.65950E-05	-2.59820E-04	4.37180E-05	-8.50976E-05	5.03628E-05	-9.56785E-05	5.03860E-05	-3.67488E-06
ANGLE	19.202500	6.28845E-04	-4.98830E-03	-1.69039E-04	-4.25002E-04	-3.53134E-04	-2.52272E-03	-3.53880E-04	-1.01962E-05
RATES		2.17231E-05	-2.59820E-04	4.72193E-05	-8.55792E-05	5.18015E-05	-8.86193E-05	5.18151E-05	-3.73260E-06
ANGLE	19.302500	6.30774E-04	-5.01428E-03	-1.64150E-04	-4.33583E-04	-3.47883E-04	-2.53121E-03	-3.48623E-04	-1.05723E-05
RATES		1.68667E-05	-2.59820E-04	5.05232E-05	-8.60416E-05	5.32366E-05	-8.12029E-05	5.32415E-05	-3.78939E-06
ANGLE	19.402500	6.32219E-04	-5.04027E-03	-1.58942E-04	-4.42209E-04	-3.42487E-04	-2.53894E-03	-3.43232E-04	-1.09541E-05
RATES		1.20308E-05	-2.59820E-04	5.35852E-05	-8.64855E-05	5.46696E-05	-7.34020E-05	5.46669E-05	-3.84527E-06
ANGLE	19.502500	6.33181E-04	-5.06625E-03	-1.53443E-04	-4.50879E-04	-3.36949E-04	-2.54587E-03	-3.47694E-04	-1.13413E-05
RATES		7.22024E-06	-2.59820E-04	5.63643E-05	-8.69114E-05	5.61017E-05	-6.51863E-05	5.60928E-05	-3.90022E-06
ANGLE	19.602500	6.33664E-04	-5.09223E-03	-1.47681E-04	-4.59590E-04	-3.31257E-04	-2.55196E-03	-3.32014E-04	-1.17341E-05
RATES		2.43962E-06	-2.59819E-04	5.88239E-05	-8.73197E-05	5.75340E-05	-5.65212E-05	5.75202E-05	-3.95424E-06
ANGLE	19.702500	6.33671E-04	-5.11821E-03	-1.41692E-04	-4.68342E-04	-3.25442E-04	-2.55715E-03	-3.26190E-04	-1.21321E-05
RATES		-2.30665E-06	-2.59819E-04	6.09321E-05	-8.77103E-05	5.89673E-05	-4.73681E-05	5.89501E-05	-4.00729E-06
ANGLE	19.802500	6.33204E-04	-5.14419E-03	-1.35511E-04	-4.77131E-04	-3.19474E-04	-2.56141E-03	-3.20224E-04	-1.25355E-05
RATES		-7.01441E-06	-2.59819E-04	6.26626E-05	-8.80831E-05	6.04020E-05	-3.76832E-05	6.03832E-05	-4.05933E-06
ANGLE	19.902500	6.32269E-04	-5.17018E-03	-1.29177E-04	-4.85958E-04	-3.13362E-04	-2.56467E-03	-3.14114E-04	-1.29439E-05
RATES		-1.16797E-05	-2.59819E-04	6.39947E-05	-8.84378E-05	6.18384E-05	-2.74168E-05	6.18198E-05	-4.11030E-06
ANGLE	20.002500	6.30870E-04	-5.19616E-03	-1.22731E-04	-4.94818E-04	-3.07106E-04	-2.56687E-03	-3.07860E-04	-1.33575E-05
RATES		-1.62990E-05	-2.59819E-04	6.49139E-05	-8.87740E-05	6.32763E-05	-1.65128E-05	6.32597E-05	-4.16013E-06
ANGLE	20.102500	6.29012E-04	-5.22214E-03	-1.16214E-04	-5.03711E-04	-3.00706E-04	-2.56794E-03	-3.01461E-04	-1.37759E-05
RATES		-2.08687E-05	-2.59818E-04	6.54119E-05	-8.90909E-05	6.47155E-05	-4.90731E-06	6.47027E-05	-4.20873E-06
ANGLE	20.202500	6.26699E-04	-5.24812E-03	-1.09669E-04	-5.12635E-04	-2.94163E-04	-2.56782E-03	-2.94919E-04	-1.41992E-05
RATES		-2.53859E-05	-2.59818E-04	6.54871E-05	-8.93879E-05	6.61552E-05	7.47242E-06	6.61479E-05	-4.25301E-06
ANGLE	20.302500	6.23937E-04	-5.27410E-03	-1.03138E-04	-5.21598E-04	-2.87475E-04	-2.56641E-03	-2.88232E-04	-1.46271E-05
RATES		-2.98478E-05	-2.59818E-04	6.51439E-05	-8.96541E-05	6.75942E-05	2.07094E-05	6.75944E-05	-4.30184E-06
ANGLE	20.402500	6.20732E-04	-5.30009E-03	-9.64915E-05	-5.30507E-04	-2.80644E-04	-2.56363E-03	-2.81400E-04	-1.50595E-05
RATES		2.60161E-05	-2.59818E-04	6.46119E-05	-8.99309E-05	6.89644E-05	-3.00000E-05	6.89644E-05	-4.34873E-06

	TIME	THETAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETAE	PHIE
ANGLE	20.502500	6.17089E-04	-5.32607E-03	-9.02300E-05	-5.39571E-04	-2.73669E-04	-2.55939E-03	-2.74423E-04	-1.54962E-05
RATES		-3.85958E-05	-2.59818E-04	6.32535E-05	-9.01518E-05	7.04645E-05	5.01480E-05	7.04850E-05	-4.38868E-06
ANGLE	20.602500	6.13016E-04	-5.35205E-03	-8.40311E-05	-5.48597E-04	-2.65551E-04	-2.55356E-03	-2.67303E-04	-1.59371E-05
RATES		-4.28777E-05	-2.59818E-04	6.17467E-05	-9.03619E-05	7.18920E-05	6.65832E-05	7.19252E-05	-4.42940E-06
ANGLE	20.702500	6.08517E-04	-5.37803E-03	-7.79501E-05	-5.57642E-04	-2.59291E-04	-2.54602E-03	-2.60039E-04	-1.63320E-05
RATES		-4.70959E-05	-2.59818E-04	5.99020E-05	-9.05490E-05	7.33116E-05	8.43475E-05	7.33589E-05	-4.46811E-06
ANGLE	20.802500	6.03600E-04	-5.40401E-03	-7.20691E-05	-5.66705E-04	-2.51889E-04	-2.53661E-03	-2.52631E-04	-1.68307E-05
RATES		-5.12490E-05	-2.59817E-04	5.77534E-05	-9.07127E-05	7.47206E-05	1.03607E-04	7.47833E-05	-4.50467E-06
ANGLE	20.902500	5.98270E-04	-5.42999E-03	-6.64163E-05	-5.75783E-04	-2.44347E-04	-2.52523E-03	-2.45082E-04	-1.72828E-05
RATES		-5.53356E-05	-2.59817E-04	5.53393E-05	-9.08530E-05	7.61162E-05	1.24553E-04	7.61957E-05	-4.53889E-06
ANGLE	21.002500	5.92536E-04	-5.45598E-03	-6.10164E-05	-5.84875E-04	-2.36666E-04	-2.51164E-03	-2.37393E-04	-1.77383E-05
RATES		-5.93548E-05	-2.59817E-04	5.27023E-05	-9.09700E-05	7.74956E-05	1.47409E-04	7.75926E-05	-4.57061E-06
ANGLE	21.102500	5.86403E-04	-5.48196E-03	-5.58892E-05	-5.93976E-04	-2.28849E-04	-2.49566E-03	-2.29564E-04	-1.81968E-05
RATES		-6.33057E-05	-2.59817E-04	4.98831E-05	-9.10640E-05	7.88555E-05	1.72435E-04	7.89709E-05	-4.59965E-06
ANGLE	21.202500	5.79878E-04	-5.50794E-03	-5.10499E-05	-6.03086E-04	-2.20896E-04	-2.47706E-03	-2.21599E-04	-1.86581E-05
RATES		-6.71877E-05	-2.59817E-04	4.69451E-05	-9.11356E-05	8.01928E-05	1.99932E-04	8.03270E-05	-4.62583E-06
ANGLE	21.302500	5.72969E-04	-5.53392E-03	-4.65090E-05	-6.12202E-04	-2.12811E-04	-2.45557E-03	-2.13500E-04	-1.91219E-05
RATES		-7.10001E-05	-2.59817E-04	4.39234E-05	-9.11852E-05	8.15041E-05	2.30256E-04	8.16573E-05	-4.64896E-06
ANGLE	21.402500	5.65681E-04	-5.55990E-03	-4.22717E-05	-6.21322E-04	-2.04596E-04	-2.43089E-03	-2.05269E-04	-1.95878E-05
RATES		-7.47425E-05	-2.59817E-04	4.08745E-05	-9.12137E-05	8.27860E-05	2.63824E-04	8.29582E-05	-4.66886E-06
ANGLE	21.502500	5.58024E-04	-5.58588E-03	-3.83380E-05	-6.30444E-04	-1.95255E-04	-2.40266E-03	-1.96909E-04	-2.00555E-05
RATES		-7.84145E-05	-2.59817E-04	3.78497E-05	-9.12216E-05	8.40352E-05	3.01127E-04	8.42261E-05	-4.68532E-06
ANGLE	21.602500	5.50002E-04	-5.61187E-03	-3.47031E-05	-6.39565E-04	-1.87791E-04	-2.37050E-03	-1.88425E-04	-2.05247E-05
RATES		-8.20155E-05	-2.59817E-04	3.49001E-05	-9.12099E-05	8.52484E-05	3.42743E-04	8.54574E-05	-4.69815E-06
ANGLE	21.702500	5.41624E-04	-5.63785E-03	-3.13568E-05	-6.48685E-04	-1.79207E-04	-2.33393E-03	-1.79820E-04	-2.09950E-05
RATES		-8.55453E-05	-2.59817E-04	3.20752E-05	-9.11791E-05	8.64222E-05	3.84930E-04	8.66487E-05	-4.70710E-06
ANGLE	21.802500	5.32896E-04	-5.66383E-03	-2.82843E-05	-6.57800E-04	-1.70508E-04	-2.29241E-03	-1.71097E-04	-2.14659E-05
RATES		-8.90034E-05	-2.59817E-04	2.94225E-05	-9.11298E-05	8.75537E-05	4.41788E-04	8.77966E-05	-4.71195E-06
ANGLE	21.902500	5.23827E-04	-5.68981E-03	-2.54600E-05	-6.66910E-04	-1.61698E-04	-2.24529E-03	-1.62262E-04	-2.19371E-05
RATES		-9.23892E-05	-2.59817E-04	2.69857E-05	-9.10625E-05	8.86399E-05	5.01013E-04	8.88981E-05	-4.71242E-06
ANGLE	22.002500	5.14622E-04	-5.71579E-03	-2.28753E-05	-6.76112E-04	-1.52789E-04	-2.19184E-03	-1.53320E-04	-2.25062E-05

	TIME	THETAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETAE	PHIE
ANGLE	22.102500	5.04689E-04	-5.74177E-03	-2.04927E-05	-6.85105E-04	-1.43765E-04	-2.13131E-03	-1.44274E-04	-2.28786E-05
RATES		-9.89418E-05	-2.59817E-04	2.29146E-05	-2.08743E-05	9.06660E-05	6.44443E-04	9.09508E-05	-4.69806E-06
ANGLE	22.202500	4.94636E-04	-5.76776E-03	-1.82816E-05	-6.94186E-04	-1.34651E-04	-2.05250E-03	-1.35132E-04	-2.33478E-05
RATES		-1.02107E-04	-2.59817E-04	2.13524E-05	-9.07536E-05	9.16010E-05	7.31575E-04	9.18967E-05	-4.68429E-06
ANGLE	22.302500	4.84270E-04	-5.79374E-03	-1.62084E-05	-7.03255E-04	-1.25446E-04	-1.98455E-03	-1.25897E-04	-2.38152E-05
RATES		-1.05197E-04	-2.59817E-04	2.01438E-05	-9.06149E-05	9.24812E-05	8.11345E-04	9.27861E-05	-4.66371E-06
ANGLE	22.402500	4.73599E-04	-5.81972E-03	-1.42392E-05	-7.12309E-04	-1.16157E-04	-1.89503E-03	-1.16576E-04	-2.42803E-05
RATES		-1.08210E-04	-2.59817E-04	1.93107E-05	-9.04582E-05	9.23051E-05	9.45802E-04	9.35172E-05	-4.63673E-06
ANGLE	22.502500	4.62631E-04	-5.84570E-03	-1.23332E-05	-7.21346E-04	-1.06787E-04	-1.79483E-03	-1.07176E-04	-2.47424E-05
RATES		-1.11146E-04	-2.59817E-04	1.88679E-05	-9.02844E-05	9.40712E-05	1.07724E-03	9.43886E-05	-4.60275E-06
ANGLE	22.602500	4.51373E-04	-5.87168E-03	-1.04517E-05	-7.30365E-04	-9.73444E-05	-1.67973E-03	-9.77006E-05	-2.52006E-05
RATES		-1.14002E-04	-2.59817E-04	1.88240E-05	-9.00953E-05	9.47787E-05	1.22801E-03	9.50994E-05	-4.56118E-06
ANGLE	22.702500	4.39833E-04	-5.89766E-03	-8.55450E-06	-7.39364E-04	-8.78136E-05	-1.54849E-03	-8.81577E-05	-2.56543E-05
RATES		-1.16778E-04	-2.59818E-04	1.91816E-05	-8.98954E-05	9.54270E-05	1.40035E-03	9.57490E-05	-4.51151E-06
ANGLE	22.802500	4.28020E-04	-5.92365E-03	-6.60173E-06	-7.48344E-04	-7.82610E-05	-1.39886E-03	-7.85529E-05	-2.61026E-05
RATES		-1.19470E-04	-2.59818E-04	1.99366E-05	-8.96937E-05	9.60159E-05	1.59585E-03	9.63371E-05	-4.45339E-06
ANGLE	22.902500	4.15943E-04	-5.94963E-03	-4.55101E-06	-7.57304E-04	-6.86329E-05	-1.22811E-03	-6.88929E-05	-2.65446E-05
RATES		-1.22077E-04	-2.59818E-04	2.10771E-05	-8.95082E-05	9.65454E-05	1.81465E-03	9.68639E-05	-4.38698E-06
ANGLE	23.002500	4.03609E-04	-5.97561E-03	-2.36769E-06	-7.66248E-04	-5.89549E-05	-1.03488E-03	-5.91832E-05	-2.69797E-05
RATES		-1.24597E-04	-2.59819E-04	2.25869E-05	-8.93656E-05	9.70162E-05	2.05426E-03	9.73300E-05	-4.31321E-06
ANGLE	23.102500	3.91028E-04	-6.00159E-03	-1.60349E-08	-7.75181E-04	-4.92326E-05	-8.16741E-04	-4.94299E-05	-2.74070E-05
RATES		-1.27026E-04	-2.59819E-04	2.44434E-05	-8.93088E-05	9.74290E-05	2.30814E-03	9.77363E-05	-4.23432E-06
ANGLE	23.202500	3.78208E-04	-6.02757E-03	2.53719E-06	-7.84117E-04	-3.94719E-05	-5.73090E-04	-3.96389E-05	-2.78264E-05
RATES		-1.29364E-04	-2.59820E-04	2.66184E-05	-8.93194E-05	9.77849E-05	2.56429E-03	9.80842E-05	-4.15426E-06
ANGLE	23.302500	3.65160E-04	-6.05356E-03	5.32212E-06	-7.93072E-04	-2.96784E-05	-3.6660E-04	-2.98159E-05	-2.82381E-05
RATES		-1.31606E-04	-2.59821E-04	2.90789E-05	-8.97162E-05	9.80854E-05	2.80394E-03	9.83750E-05	-4.07907E-06
ANGLE	23.402500	3.51842E-04	-6.07954E-03	8.36541E-06	-8.02075E-04	-1.98575E-05	-1.43040E-05	-1.99666E-05	-2.86429E-05
RATES		-1.33750E-04	-2.59821E-04	3.17369E-05	-9.03457E-05	9.83320E-05	3.00316E-03	9.86108E-05	-4.01638E-06
ANGLE	23.502500	3.38414E-04	-6.10552E-03	1.16873E-05	-8.11156E-04	-1.00140E-05	2.93539E-04	-1.00958E-05	-2.90421E-05
RATES		-1.35793E-04	-2.59822E-04	3.47061E-05	-9.15539E-05	9.85264E-05	3.13988E-03	9.87932E-05	-3.97325E-06
ANGLE	23.602500	3.24736E-04	-6.13150E-03	1.53105E-05	-8.20360E-04	-1.53670E-07	6.10726E-04	-2.07430E-07	-2.94311E-05

	TIME	THETAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETA E	PHIE
ANGLE RATES	23.702500	3.10871E-04 -1.39565E-04	-6.15749E-03 -2.59824E-04	1.92475E-05 4.09747E-05	-8.29720E-04 -9.46356E-05	9.71870E-06 9.87683E-05	9.28181E-04 3.14401E-03	9.68765E-06 9.90085E-05	-2.98347E-05 -3.97161E-06
ANGLE RATES	23.802500	2.96827E-04 -1.41287E-04	-6.18347E-03 -2.59825E-04	2.35070E-05 4.42267E-05	-8.39296E-04 -9.67922E-05	1.95985E-05 9.88204E-05	1.23669E-03 3.01138E-03	1.95908E-05 9.90464E-05	-3.02337E-05 -4.01315E-06
ANGLE RATES	23.902500	2.82617E-04 -1.42876E-04	-6.20945E-03 -2.59825E-04	2.80930E-05 4.74885E-05	-8.49092E-04 -9.91328E-05	2.94814E-05 9.88298E-05	1.52829E-03 2.81087E-03	2.94955E-05 9.90413E-05	-3.00380E-05 -4.07600E-06
ANGLE RATES	24.002500	2.68252E-04 -1.44389E-04	-6.23543E-03 -2.59826E-04	3.30034E-05 5.07079E-05	-8.59124E-04 -1.01512E-04	3.93631E-05 9.87993E-05	1.79755E-03 2.56994E-03	3.93977E-05 9.89960E-05	-3.10493E-05 -4.15147E-06
ANGLE RATES	24.102500	2.53743E-04 -1.45764E-04	-6.26142E-03 -2.59826E-04	3.82315E-05 5.38332E-05	-8.69392E-04 -1.03814E-04	4.92400E-05 9.87312E-05	2.04173E-03 2.31329E-03	4.92935E-05 9.89134E-05	-3.14684E-05 -4.23170E-06
ANGLE RATES	24.202500	2.39103E-04 -1.47016E-04	-6.28740E-03 -2.59827E-04	4.37653E-05 5.68140E-05	-8.79882E-04 -1.05958E-04	5.91082E-05 9.86281E-05	2.26027E-03 2.05987E-03	5.91792E-05 9.87961E-05	-3.18956E-05 -4.31064E-06
ANGLE RATES	24.302500	2.24344E-04 -1.48145E-04	-6.31338E-03 -2.59827E-04	4.95879E-05 5.96023E-05	-8.90577E-04 -1.07901E-04	6.89645E-05 9.84924E-05	2.45418E-03 1.82191E-03	6.90516E-05 9.86468E-05	-3.23304E-05 -4.38434E-06
ANGLE RATES	24.402500	2.09478E-04 -1.49146E-04	-6.33936E-03 -2.59827E-04	5.56778E-05 6.21527E-05	-9.01455E-04 -1.09626E-04	7.88057E-05 9.83263E-05	2.62536E-03 1.60584E-03	7.89076E-05 9.84682E-05	-3.27722E-05 -4.45068E-06
ANGLE RATES	24.502500	1.94519E-04 -1.50019E-04	-6.36535E-03 -2.59827E-04	6.20091E-05 6.44235E-05	-9.12495E-04 -1.11136E-04	8.85288E-05 9.81320E-05	2.77614E-03 1.41392E-03	8.87443E-05 9.82625E-05	-3.32203E-05 -4.50885E-06
ANGLE RATES	24.602500	1.79479E-04 -1.50762E-04	-6.39133E-03 -2.59827E-04	6.85520E-05 6.63773E-05	-9.23676E-04 -1.12443E-04	9.84312E-05 9.79114E-05	2.90893E-03 1.24583E-03	9.85593E-05 9.80320E-05	-3.36737E-05 -4.55886E-06
ANGLE RATES	24.702500	1.64371E-04 -1.51372E-04	-6.41731E-03 -2.59828E-04	7.52729E-05 6.79812E-05	-9.34977E-04 -1.13562E-04	1.08210E-04 9.76662E-05	3.02604E-03 1.09987E-03	1.08350E-04 9.77786E-05	-3.41318E-05 -4.60116E-06
ANGLE RATES	24.802500	1.49209E-04 -1.51848E-04	-6.44329E-03 -2.59828E-04	8.21356E-05 6.92078E-05	-9.46382E-04 -1.14512E-04	1.17964E-04 9.73978E-05	3.12956E-03 9.73718E-04	1.18114E-04 9.75037E-05	-3.45927E-05 -4.63639E-06
ANGLE RATES	24.902500	1.34006E-04 -1.52189E-04	-6.46928E-03 -2.59828E-04	8.91011E-05 7.00355E-05	-9.57875E-04 -1.15307E-04	1.27689E-04 9.71074E-05	3.22133E-03 8.64928E-04	1.27850E-04 9.72086E-05	-3.50589E-05 -4.66525E-06
ANGLE RATES	25.002500	1.18776E-04 -1.52395E-04	-6.49526E-03 -2.59828E-04	9.61287E-05 7.04484E-05	-9.69439E-04 -1.15962E-04	1.37384E-04 9.67958E-05	3.30300E-03 7.71114E-04	1.37555E-04 9.68942E-05	-3.55266E-05 -4.68842E-06
ANGLE RATES	25.102500	1.03531E-04 -1.52465E-04	-6.52124E-03 -2.59828E-04	1.03176E-04 7.04369E-05	-9.81063E-04 -1.16489E-04	1.47048E-04 9.64636E-05	3.37594E-03 6.90157E-04	1.47228E-04 9.65610E-05	-3.59969E-05 -4.70648E-06
ANGLE RATES	25.202500	8.82870E-05	-6.54723E-03	1.10201E-04	-9.92734E-04	1.56677E-04	3.44136E-03	1.56867E-04	-3.64679E-05

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ANGLE	25.302500	7.30560E-05	-6.57321E-03	1.17161E-04	-1.00444E-03	1.66269E-04	3.50026E-03	1.66470E-04	-3.69402E-05
RATES		-1.52198E-04	-2.59828E-04	6.91360E-05	-1.17193E-04	9.57381E-05	5.59568E-04	9.58387E-05	-4.72925E-06
ANGLE	25.402500	5.78519E-05	-6.59919E-03	1.24014E-04	-1.01617E-03	1.75823E-04	3.55351E-03	1.76034E-04	-3.74135E-05
RATES		-1.51862E-04	-2.59828E-04	6.78598E-05	-1.17383E-04	9.53444E-05	5.06953E-04	9.54489E-05	-4.73471E-06
ANGLE	25.502500	4.26881E-05	-6.62517E-03	1.30720E-04	-1.02791E-03	1.85357E-04	3.60185E-03	1.85559E-04	-3.78871E-05
RATES		-1.51392E-04	-2.59828E-04	6.61861E-05	-1.17470E-04	9.49291E-05	4.61174E-04	9.50392E-05	-4.73660E-06
ANGLE	25.602500	2.75780E-05	-6.65116E-03	1.37239E-04	-1.03966E-03	1.94609E-04	3.64591E-03	1.95041E-04	-3.83607E-05
RATES		-1.50789E-04	-2.59828E-04	6.41370E-05	-1.17459E-04	9.44914E-05	4.21255E-04	9.46084E-05	-4.73510E-06
ANGLE	25.702500	1.25348E-05	-6.67714E-03	1.43535E-04	-1.05140E-03	2.04235E-04	3.68425E-03	2.04480E-04	-3.88340E-05
RATES		-1.50055E-04	-2.59828E-04	6.17404E-05	-1.17350E-04	9.40300E-05	3.94377E-04	9.41553E-05	-4.73035E-06
ANGLE	25.802500	-2.42865E-06	-6.70312E-03	1.49576E-04	-1.06313E-03	2.13614E-04	3.72332E-03	2.13872E-04	-3.93067E-05
RATES		-1.49192E-04	-2.59828E-04	5.90295E-05	-1.17144E-04	9.35434E-05	3.55848E-04	9.36782E-05	-4.72242E-06
ANGLE	25.902500	-1.72994E-05	-6.72910E-03	1.55332E-04	-1.07483E-03	2.22943E-04	3.75753E-03	2.23214E-04	-3.97784E-05
RATES		-1.48202E-04	-2.59828E-04	5.60423E-05	-1.16341E-04	9.30298E-05	3.29086E-04	9.31753E-05	-4.71132E-06
ANGLE	26.002500	-3.20649E-05	-6.75509E-03	1.60777E-04	-1.08649E-03	2.32219E-04	3.78923E-03	2.32506E-04	-4.02488E-05
RATES		-1.47087E-04	-2.59828E-04	5.28206E-05	-1.16442E-04	9.24873E-05	3.05601E-04	9.26446E-05	-4.69703E-06
ANGLE	26.102500	-4.67128E-05	-6.78107E-03	1.65890E-04	-1.09811E-03	2.41439E-04	3.81873E-03	2.41742E-04	-4.07177E-05
RATES		-1.45851E-04	-2.59828E-04	4.94101E-05	-1.15945E-04	9.19139E-05	2.84977E-04	9.20838E-05	-4.67948E-06
ANGLE	26.202500	-6.12311E-05	-6.80705E-03	1.70654E-04	-1.10968E-03	2.50600E-04	3.84630E-03	2.50921E-04	-4.11846E-05
RATES		-1.44496E-04	-2.59828E-04	4.58590E-05	-1.15350E-04	9.13073E-05	2.66862E-04	9.14906E-05	-4.65857E-06
ANGLE	26.302500	-7.56080E-05	-6.83304E-03	1.75059E-04	-1.12118E-03	2.59699E-04	3.87217E-03	2.60039E-04	-4.16493E-05
RATES		-1.43024E-04	-2.59828E-04	4.22177E-05	-1.14656E-04	9.06652E-05	2.50955E-04	9.08622E-05	-4.63417E-06
ANGLE	26.402500	-8.98322E-05	-6.85902E-03	1.79097E-04	-1.13261E-03	2.68732E-04	3.89655E-03	2.69092E-04	-4.21113E-05
RATES		-1.41439E-04	-2.59828E-04	3.85376E-05	-1.13860E-04	8.92953E-05	2.37001E-04	9.01962E-05	-4.60610E-06
ANGLE	26.502500	-1.03892E-04	-6.88500E-03	1.82767E-04	-1.14395E-03	2.77595E-04	3.91962E-03	2.78077E-04	-4.25704E-05
RATES		-1.39745E-04	-2.59828E-04	3.48707E-05	-1.12964E-04	8.82653E-05	2.24779E-04	8.94898E-05	-4.57419E-06
ANGLE	26.602500	-1.17778E-04	-6.91098E-03	1.86073E-04	-1.15519E-03	2.86584E-04	3.94155E-03	2.86989E-04	-4.30260E-05
RATES		-1.37945E-04	-2.59828E-04	3.12686E-05	-1.11965E-04	8.85029E-05	2.14103E-04	8.87405E-05	-4.53823E-06
ANGLE	26.702500	-1.31478E-04	-6.93697E-03	1.89025E-04	-1.16634E-03	2.95394E-04	3.96249E-03	2.95324E-04	-4.34779E-05
RATES		-1.36041E-04	-2.59828E-04	2.77816E-05	-1.10863E-04	8.76958E-05	2.04910E-04	8.77458E-05	-4.49790E-06
ANGLE	26.802500	-1.44983E-04	-6.96295E-03	1.91636E-04	-1.17736E-03	3.04121E-04	3.98255E-03	3.04577E-04	-4.39250E-05

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	TIME	THETAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETA	PHIE
ANGLE RATES	26.902500	-2.58282E-04 -1.31936E-04	-6.98893E-03 -2.59828E-04	1.93924E-04 2.13440E-05	-1.18826E-03 -1.08347E-04	3.12761E-04 8.59395E-05	4.00187E-03 1.89829E-04	3.13243E-04 8.62107E-05	-4.43684E-05 -4.40374E-06
ANGLE RATES	27.002500	-1.71367E-04 -1.29742E-04	-7.01492E-03 -2.59828E-04	1.95913E-04 1.84813E-05	-1.19903E-03 -1.06933E-04	3.21308E-04 8.49867E-05	4.02055E-03 1.83915E-04	3.21817E-04 8.52664E-05	-4.43061E-05 -4.34926E-06
ANGLE RATES	27.102500	-1.84227E-04 -1.27458E-04	-7.04090E-03 -2.59828E-04	1.97631E-04 1.59081E-05	-1.20965E-03 -1.05415E-04	3.29756E-04 8.39820E-05	4.03868E-03 1.74924E-04	3.30294E-04 8.42685E-05	-4.52380E-05 -4.28954E-06
ANGLE RATES	27.202500	-1.96855E-04 -1.25086E-04	-7.06688E-03 -2.59828E-04	1.99106E-04 1.36579E-05	-1.22011E-03 -1.03794E-04	3.38102E-04 8.29243E-05	4.05636E-03 1.74775E-04	3.38669E-04 8.32159E-05	-4.56638E-05 -4.22436E-06
ANGLE RATES	27.302500	-2.09242E-04 -1.22630E-04	-7.09286E-03 -2.59828E-04	2.00375E-04 1.17588E-05	-1.23040E-03 -1.02070E-04	3.46339E-04 8.18126E-05	4.07366E-03 1.71398E-04	3.46935E-04 8.21074E-05	-4.60827E-05 -4.15349E-06
ANGLE RATES	27.402500	-2.21379E-04 -1.20091E-04	-7.11885E-03 -2.59828E-04	2.01471E-04 1.02335E-05	-1.24052E-03 -1.00243E-04	3.54463E-04 8.05462E-05	4.09066E-03 1.68729E-04	3.55088E-04 8.09426E-05	-4.64943E-05 -4.07674E-06
ANGLE RATES	27.502500	-2.33258E-04 -1.17474E-04	-7.14483E-03 -2.59828E-04	2.02435E-04 9.09829E-06	-1.25045E-03 -9.83143E-05	3.62467E-04 7.94249E-05	4.10743E-03 1.66715E-04	3.63122E-04 7.97211E-05	-4.68979E-05 -3.99391E-06
ANGLE RATES	27.602500	-2.44871E-04 -1.14778E-04	-7.17081E-03 -2.59828E-04	2.03305E-04 8.36355E-06	-1.26018E-03 -9.62850E-05	3.70346E-04 7.81488E-05	4.12403E-03 1.65305E-04	3.71031E-04 7.84429E-05	-4.72929E-05 -3.90486E-06
ANGLE RATES	27.702500	-2.56210E-04 -1.12008E-04	-7.19679E-03 -2.59828E-04	2.04125E-04 8.03367E-06	-1.26970E-03 -9.41556E-05	3.78094E-04 7.68184E-05	4.14051E-03 1.64452E-04	3.78808E-04 7.71084E-05	-4.76786E-05 -3.80942E-06
ANGLE RATES	27.802500	-2.67269E-04 -1.09165E-04	-7.22278E-03 -2.59828E-04	2.04932E-04 8.10595E-06	-1.27901E-03 -9.19267E-05	3.85707E-04 7.54343E-05	4.15694E-03 1.64121E-04	3.86449E-04 7.57184E-05	-4.80544E-05 -3.70751E-06
ANGLE RATES	27.902500	-2.78039E-04 -1.06250E-04	-7.24876E-03 -2.59827E-04	2.05766E-04 8.57044E-06	-1.28808E-03 -8.95988E-05	3.93178E-04 7.39977E-05	4.17337E-03 1.64276E-04	3.93949E-04 7.42740E-05	-4.84197E-05 -3.59905E-06
ANGLE RATES	28.002500	-2.88515E-04 -1.03266E-04	-7.27474E-03 -2.59827E-04	2.06666E-04 9.41082E-06	-1.29692E-03 -8.71722E-05	4.00504E-04 7.25101E-05	4.18982E-03 1.64887E-04	4.01302E-04 7.27768E-05	-4.87739E-05 -3.48403E-06
ANGLE RATES	28.102500	-2.98689E-04 -1.00212E-04	-7.30073E-03 -2.59827E-04	2.07668E-04 1.06056E-05	-1.30551E-03 -8.46457E-05	4.07678E-04 7.09735E-05	4.20637E-03 1.65923E-04	4.08502E-04 7.12286E-05	-4.91162E-05 -3.36243E-06
ANGLE RATES	28.202500	-3.08554E-04 -9.73908E-05	-7.32671E-03 -2.59827E-04	2.08306E-04 1.21277E-05	-1.31384E-03 -8.20223E-05	4.14696E-04 6.93898E-05	4.22303E-03 1.67361E-04	4.15545E-04 6.96316E-05	-4.94461E-05 -3.23430E-06
ANGLE RATES	28.302500	-3.18104E-04 -9.39029E-05	-7.35269E-03 -2.59827E-04	2.10112E-04 1.39446E-05	-1.32191E-03 -7.92982E-05	4.21553E-04 6.77616E-05	4.23986E-03 1.69174E-04	4.22425E-04 6.79884E-05	-4.97628E-05 -3.09972E-06
ANGLE RATES	28.402500	-3.27331E-04	-7.37867E-03	2.11612E-04	-1.32970E-03	4.28246E-04	4.25689E-03	4.29140E-04	-5.00657E-05

	TIME	THETAB	PHIB	THETAS	PHIS	THETAP	PHIP	THETA E	PHIE
ANGLE	28.502500	-3.36230E-04	-7.40466E-03	2.13330E-04	-1.33720E-03	4.34769E-04	4.27415E-03	4.35684E-04	-5.03542E-05
RATES		-8.73304E-05	-2.59827E-04	1.83104E-05	-7.35473E-05	6.43823E-05	1.73847E-04	6.45750E-05	-2.81165E-06
ANGLE	28.602500	-3.44794E-04	-7.43064E-03	2.15287E-04	-1.34440E-03	4.41120E-04	4.29167E-03	4.42053E-04	-5.06277E-05
RATES		-8.39471E-05	-2.59828E-04	2.07730E-05	-7.05176E-05	6.26374E-05	1.76668E-04	6.28113E-05	-2.65850E-06
ANGLE	28.702500	-3.53016E-04	-7.45662E-03	2.17496E-04	-1.35130E-03	4.47295E-04	4.30950E-03	4.48244E-04	-5.08856E-05
RATES		-8.04999E-05	-2.59828E-04	2.33597E-05	-6.73829E-05	6.08598E-05	1.79790E-04	6.10144E-05	-2.49955E-06
ANGLE	28.802500	-3.60891E-04	-7.48260E-03	2.19967E-04	-1.35788E-03	4.53290E-04	4.32765E-03	4.54254E-04	-5.11273E-05
RATES		-7.69894E-05	-2.59828E-04	2.60207E-05	-6.41411E-05	5.90530E-05	1.83197E-04	5.91879E-05	-2.33563E-06
ANGLE	28.902500	-3.68411E-04	-7.50859E-03	2.22706E-04	-1.36412E-03	4.59104E-04	4.34615E-03	4.60080E-04	-5.13523E-05
RATES		-7.34160E-05	-2.59828E-04	2.87052E-05	-6.07900E-05	5.72205E-05	1.86876E-04	5.73355E-05	-2.16521E-06
ANGLE	29.002500	-3.75571E-04	-7.53457E-03	2.25712E-04	-1.37003E-03	4.64733E-04	4.36504E-03	4.65719E-04	-5.15601E-05
RATES		-6.97804E-05	-2.59828E-04	3.13618E-05	-5.73273E-05	5.53658E-05	1.90813E-04	5.54611E-05	-1.99039E-06
ANGLE	29.102500	-3.82364E-04	-7.56055E-03	2.28979E-04	-1.37558E-03	4.70176E-04	4.38433E-03	4.71171E-04	-5.17501E-05
RATES		-6.60830E-05	-2.59828E-04	3.39393E-05	-5.37508E-05	5.34925E-05	1.94998E-04	5.35685E-05	-1.81086E-06
ANGLE	29.202500	-3.88784E-04	-7.58654E-03	2.32498E-04	-1.38077E-03	4.75430E-04	4.40405E-03	4.75432E-04	-5.19220E-05
RATES		-6.23247E-05	-2.59828E-04	3.63878E-05	-5.00581E-05	5.16038E-05	1.99420E-04	5.16612E-05	-1.62695E-06
ANGLE	29.302500	-3.94826E-04	-7.61252E-03	2.36253E-04	-1.38559E-03	4.80496E-04	4.42423E-03	4.81502E-04	-5.20753E-05
RATES		-5.85062E-05	-2.59828E-04	3.86594E-05	-4.62472E-05	4.97033E-05	2.04069E-04	4.97428E-05	-1.43897E-06
ANGLE	29.402500	-4.00483E-04	-7.63850E-03	2.40224E-04	-1.39002E-03	4.85370E-04	4.44488E-03	4.85380E-04	-5.22096E-05
RATES		-5.46284E-05	-2.59828E-04	4.07087E-05	-4.23162E-05	4.77942E-05	2.08937E-04	4.78168E-05	-1.24726E-06
ANGLE	29.502500	-4.05749E-04	-7.66448E-03	2.44386E-04	-1.39404E-03	4.90054E-04	4.46602E-03	4.91065E-04	-5.23245E-05
RATES		-5.06925E-05	-2.59828E-04	4.24938E-05	-3.82634E-05	4.58795E-05	2.14017E-04	4.58863E-05	-1.05213E-06
ANGLE	29.602500	-4.10618E-04	-7.69047E-03	2.48711E-04	-1.39766E-03	4.94546E-04	4.48769E-03	4.95557E-04	-5.24198E-05
RATES		-4.66998E-05	-2.59828E-04	4.39766E-05	-3.40876E-05	4.39621E-05	2.19301E-04	4.39544E-05	-8.53916E-07
ANGLE	29.702500	-4.15096E-04	-7.71645E-03	2.53163E-04	-1.40086E-03	4.98846E-04	4.50989E-03	4.99855E-04	-5.24951E-05
RATES		-4.26519E-05	-2.59828E-04	4.51240E-05	-2.97878E-05	4.20447E-05	2.24785E-04	4.20239E-05	-6.52912E-07
ANGLE	29.802500	-4.19146E-04	-7.74243E-03	2.57721E-04	-1.40361E-03	5.02955E-04	4.53266E-03	5.03961E-04	-5.25502E-05
RATES		-3.85505E-05	-2.59828E-04	4.59078E-05	-2.53638E-05	4.01297E-05	2.30464E-04	4.00971E-05	-4.49408E-07
ANGLE	29.902500	-4.22793E-04	-7.76842E-03	2.62333E-04	-1.40592E-03	5.06872E-04	4.55600E-03	5.07875E-04	-5.25849E-05
RATES		-3.43976E-05	-2.59828E-04	4.63055E-05	-2.08157E-05	3.82191E-05	2.36333E-04	3.81765E-05	-2.43671E-07
ANGLE	30.002500	-4.26023E-04	-7.79440E-03	2.66964E-04	-1.40777E-03	5.10599E-04	4.57993E-03	5.11597E-04	-5.25988E-05

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46	-2.265785019396852E-03
47	-2.405877807910656E-03
48	-2.550956197100496E-03
49	-2.701038762360677E-03
50	-2.856139214816519E-03
51	-3.016266360423980E-03
52	-3.181424062399962E-03
53	-3.351611207027247E-03
54	-3.526821672852343E-03
55	-3.707044303313881E-03
56	-3.892262882817382E-03
57	-4.082450116288352E-03
58	-4.277597612214162E-03
59	-4.477655859206343E-03
60	-4.682594266087831E-03
61	-4.892371055531336E-03
62	-5.106939361249008E-03
63	-5.326247178758126E-03
64	-5.550237379716216E-03
65	-5.778847719847213E-03
66	-6.012010850446725E-03
67	-6.249654333483429E-03
68	-6.491700660283234E-03
69	-6.738067273804438E-03
70	-6.988666594492488E-03
71	-7.243406049708967E-03
72	-7.502188106727314E-03
73	-7.764910309285278E-03
74	-8.031465317673403E-03
75	-8.301740952356751E-03
76	-8.575620241096092E-03
77	-8.852981469575638E-03
78	-9.133698235462839E-03
79	-9.417639505953423E-03
80	-9.704669678701927E-03
81	-9.994048646155124E-03
82	-1.028743186323750E-02
83	-1.058287041836525E-02
84	-1.088001110774719E-02
85	-1.118109651294946E-02
86	-1.148356508166282E-02
87	-1.178805121165843E-02
88	-1.209430533786689E-02
89	-1.240239402255133E-02
90	-1.271190004851896E-02
91	-1.302272251532700E-02
92	-1.333467093843615E-02
93	-1.364757535124389E-02
94	-1.396122020796681E-02

97	-1.490413509589096E-02
98	-1.521941813161382E-02
99	-1.553385174870464E-02
100	-1.584702524431705E-02
101	-1.616112679708392E-02
102	-1.647354214253494E-02
103	-1.678485470078261E-02
104	-1.709484570641506E-02
105	-1.740329434051224E-02
106	-1.770997786472539E-02
107	-1.801467175732763E-02
108	-1.831714985118000E-02
109	-1.861718447351557E-02
110	-1.891454058746983E-02
111	-1.920900593528474E-02
112	-1.950033118309413E-02
113	-1.978829006720872E-02
114	-2.007264954182830E-02
115	-2.035317592807517E-02
116	-2.062963506428983E-02
117	-2.090179245747636E-02
118	-2.116941343582635E-02
119	-2.14322633022370E-02
120	-2.169010748865441E-02
121	-2.194271171140283E-02
122	-2.218904212698277E-02
123	-2.243126548867025E-02
124	-2.266674930358714E-02
125	-2.289606199019822E-02
126	-2.311897303617372E-02
127	-2.333525315648277E-02
128	-2.354467445165842E-02
129	-2.374701056610484E-02
130	-2.394203684638974E-02
131	-2.412953049938726E-02
132	-2.430927075021205E-02
133	-2.448103899981824E-02
134	-2.464401898219494E-02
135	-2.479979692103240E-02
136	-2.494636108579889E-02
137	-2.508410494709268E-02
138	-2.521282133121394E-02
139	-2.533230857387129E-02
140	-2.544236747261876E-02
141	-2.554280303892920E-02
142	-2.563342264812265E-02
143	-2.571403818872997E-02
144	-2.578446521022593E-02
145	-2.584452326731397E-02

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148

-2.596074237586188E-02

149

-2.597760532807880E-02

150

-2.598326221657343E-02

PROGRAM : ACATE(INPUT,OUTPUT,TAPE5=INPUT,TAPE6=OUTPUT,PUNCH)

C
C
C****
C
C
C*
C*
C*****
C*
C* PURPOSE
C*
C* PROGRAM LACATE INTEGRATES THE LAGRANGIAN EQUATIONS OF MOTION
C* FOR THE NASA-LRC/AFCL LACATE BALLOON SYSTEM IN RESPONSE TO
C* A SUDDEN WIND GUST. THE PENDULATION MOTION IS HANDLED IN THREE
C* DIMENSIONS BY SUBROUTINE PEND. TORSIONAL MOTION IS CONSIDERED
C* IN TWO DIMENSIONS BY SUBROUTINE TORS. SUBROUTINE CORL TAKES
C* INTO ACCOUNT CORIOLIS FORCE EFFECTS IN TWO DIMENSIONS. THE
C* GUST LOADING MODEL DESIRED BY THE USER IS CHOSEN THROUGH USE
C* OF SUBROUTINE GUST. THE SMALL ANGLE APPROXIMATION FOR THE
C* EQUATIONS OF MOTION IS EMPLOYED THROUGHOUT. SUBROUTINE PLOTS
C* PROVIDES PLOTS OF THE SOLUTIONS TO THE EQUATIONS OF MOTION
C* (ANGULAR VELOCITIES AND ANGLES). DAMPING MAY ALSO BE INCLUDED.
C*
C* PROPRIETARY
C*
C* NONE
C*
C* USE
C*
C* CALL LACATE
C*
C* PARAMETERS
C*
C* NONE
C*
C* REQUIRED ROUTINES
C*
C* GUST,DIFSUB,DIFFUN,PEND,TORS,CORL
C*
C* AUTHOR/IMPLEMENTOR
C*

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C* D.R. DORSFY, JR
C* COMPUTER SCIENCES CORPORATION -
C* NASA/LANGLEY RESEARCH CENTER PROGRAM OFFICE

C* DATE RELEASED
C* SEPTEMBER 14, 1973

C* LATEST REVISION
C* SEPTEMBER 14, 1973

C* *****
C* *****

BBBBBBBBBBBBBB

---L A C A T E---

.....BALLOON

RADIUS = 64.008 M
WEIGHT = 861.825 KG
VOLUME = 39M CUFT

...HELIUM FILLED
...RADAR REFLECTOR
...AFCLR PARACHUTE

A
A

C	A			
C	A			
C	A		A = 23.1648 M	
C	A			
C	A			
C	XXXXX			
C	* X X	*LOAD BAR	
C	* X X	*		
C	* X X	*	RADIUS = 0.6096 M	
C	X X		LENGTH = 2.1336 M	
C	XXXXX		WEIGHT = 112.9445 KG	
C	L			
C	L			
C	L		...FLIGHT CONTROL	
C	L		ELECTRONICS	
C	L		...DIPOLE ANTENNAS	
C	L		...REEL-DOWN	
C	L			
C	L			
C	L			
C	L			
C	L		L = 396.24 M	
C	L			
C	L			
C	L			
C	L			
C	HHHHH			
C	H H	PARACHUTE PACKAGE	
C	H H			
C	H H		RADIUS = 0.4572 M	
C	H H		LENGTH = 1.5240 M	
C	HHHHH		WEIGHT = 37.1946 KG	
C	P			
C	P		...AFCLR ELECTRONICS 2	
C	P		...GONDOLA PARACHUTE	
C	P			
C	P		P = 3.3528 M	
C	P			
C	SSSSS	PAYLOAD PACKAGE	
C	S S			
C	SSSSS		RADIUS = 0.9144 M	
C	RRR		LENGTH = 1.8288 M	

R R
R R R
C C C C C C C C
C C
C C
C C C

WEIGHT = 214.0956 KG
...RADIOMETER (NASA)
...INSTRUMENTATION
...CRUSH PAD

** L A C A T E **

RB

THETAB,PHIB

FOUR
COUPLED
PENDULUM
CONFIGURATION

DEFINITION OF ANGLES
AND LENGTHS

A1

THETAS,PHIS

PHIP

THETAP,PHIP

```

C      Z      I
C      +
C      I+
C      I +
C      I + P
C      I +
C      I* **      THETA,PHI
C      I ** +
C      I +
C      I +
C      I , 0
C
C
C
C*****

```

```

C
000003      COMMON / CONST / MB,MS,MP,ME,M,RB,A1,L,P,RS,RP,RE,G,RHO,MU,N,
*           HS,HP,HE
000003      COMMON / CORID / OMEGA,LAMDA
000003      COMMON / DAMPC / ASCS,APCP,AECE,D(8),IDAMP
000003      COMMON / GUSTA / TONB,DOTB
000003      COMMON / TESTC / IPEND,ITORS,ICURL
000003      REAL MB,MS,MP,ME,M,L,MU,LAMDA
000003      DIMENSION Y(128),DY(16),WK(48)
000003      DIMENSION M(4)

```

```

C*****
000003      NAMELIST / ANGL / THETA,THETAS,THETAP,THETA,PHIB,PHIS,PHIP,PHIE,
1           TDOTS,TDOTP,TDOTE,DOTS,DOTP,DOTE,
2           OMEGA,LAMDA
000003      NAMELIST / CONS / MB,MS,MP,ME,M,RB,A1,L,P,RS,RP,RE,G,RHO,MU,N,
*           HS,HP,HE
000003      NAMELIST / TEST / IPEND,ITORS,ICURL,IPLT,IDAMP
000003      NAMELIST / DIF1 / H,JSTART,HMIN,HMAX,EPS,MTH,MAXDER,TO,TF,PV
000003      NAMELIST / DRAG / ASCS,APCP,AECE
C*****

```

```

C
C      READ NAMELIST INPUTS
C
000003      READ(5,ANGL)
000006      READ(5,CONS)
000011      READ(5,TEST)
000014      READ(5,DIF1)
000017      READ(5,DRAG)
C

```

```

----- C----- WRITE NAMELIST INPUTS
----- E
000022     WRITE(6,ANGL)
000025     WRITE(6,CONS)
000030     WRITE(6,TEST)
000033     WRITE(6,DIF1)
000036     WRITE(6,DRAG)
----- C
----- C--- CHOOSE DESIRED GUST MODEL AND SET UP INITIAL CONDITIONS FOR
----- C--- BALLOON VELOCITIES
----- C
000041     CALL GUST
----- C
----- C   INITIALIZE THE VALUES FOR DIFSUB
----- C
000042     T=T0
000043     N1=N*2
000045     LNCNT=17
----- C
----- C   SET Y VALUES AS INPUTS FOR DIFSUB
----- C
000046     Y(1)=THETA0
000047     Y(2)=PHI0
000051     Y(3)=THETA5
000052     Y(4)=PHI5
000054     Y(5)=THETA10
000055     Y(6)=PHI10
000057     Y(7)=THETA15
000060     Y(8)=PHI15
000062     Y(9)=TDOT0
000063     Y(10)=DDOT0
000065     Y(11)=TDOT5
000066     Y(12)=DDOT5
000070     Y(13)=TDOT10
000071     Y(14)=DDOT10
000073     Y(15)=TDOT15
000074     Y(16)=DDOT15
----- C
----- C
----- C   INTEGRATE EQUATIONS OF MOTION
----- C
000076     20 CALL DIFSUB(N1,T,H,Y,HK,JSTART,HMIN,HMAX,EPS,MTH,MAXDER,IERR)
----- C

```

```

C CHECK IF THE STEP WAS SUCCESSFUL
C
000112 IF(IERR.EQ.1) GO TO 50
C
C PRINT THE FOLLOWING IF THE STEP WAS UNSUCCESSFUL
C
000114 WRITE(6,30) IERR
000122 30 FORMAT(1H0//5X,5HIERR=,I2//2X,3HWK=)
000122 WRITE(6,40) (WK(I),I=1,112)
000134 40 FORMAT(1H0,5X,4E29.15)
000134 RETURN
C
C IF THE STEP WAS SUCCESSFUL THEN WRITE THE FOLLOWING
C
000136 50 IF(LNCNT.LT.16) GO TO 60
000141 WRITE(6,55)
000144 55 FORMAT(1H1,11X,4HTIME,9X,6HTHETAB,9X,4HPHIB,9X,6HTHETAS,9X,4HPHIS,
1 9X,6HTHETAP,9X,4HPHIP,9X,6HTHETAE,9X,4HPHIE/)
000144 LNCNT=0
000145 60 WRITE(6,65) T,(Y(I),I=1,N1)
000162 65 FORMAT(6H ANGLE,F11.6,3X,8(2X,E12.5)/6H RATES,14X,8(2X,E12.5)/)
000162 PUNCH 68 .T,(Y(I),I=1,8)
000176 PUNCH 68 .T,(Y(I),I=9,16)
000212 68 FORMAT(1X,F7.3,8E9.2)
000212 LNCNT=LNCNT+1
C
C CHECK IF THE TIME LIMIT HAS BEEN EXCEEDED
C
000214 IF(L.LI.TF) GO TO 20
000216 WRITE(6,70) H,JSTART,IERR
000230 70 FORMAT(///,5X,E25.15,2(10X,(2))
C
C SET UP PLOTS IF DESIRED
C
000230 IF(I.PLOT.EQ.1) CALL PLOTS
C
C
000233 RETURN
C
C
000235 END

```


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SUBROUTINE GUST

C
C*****
C*
C* PURPOSE
C*
C* SUBROUTINE GUST COMPUTES THE ANGULAR ACCELERATION PRODUCED ON
C* THE BALLOON BY A GIVEN GUST VELOCITY PROFILE. THE VELOCITY
C* PROFILE MAY BE CHOSEN BY THE USER FROM SUBROUTINE GUST. THE
C* FINAL VALUE OF THE W ARRAY IS TAKEN AS THE INITIAL CONDITION
C* ON ANGULAR VELOCITY FOR THE BALLOON.
C* THE ROUTINE ALSO DEFINES THE EQUATION FOR THE GUST
C* VELOCITY PROFILE. THE ARRAY AA RETURNS THE VELOCITIES FOR
C* -RB.LE.X.LE.O., WHILE ARRAY BB CONTAINS VELOCITIES FOR THE
C* RANGE O.LE.X.LE.+RB.
C*
C* PROPRIETARY
C*
C* NONE
C*
C* USE
C*
C* CALL GUST
C*
C* PARAMETERS
C*
C* NONE
C*
C* REQUIRED ROUTINES
C*
C* NONE
C*
C* AUTHOR/IMPLEMENTOR
C*
C* D.R.DORSEY,JR - B.SEAY
C*
C* DATE RELEASED
C*
C* SEPTEMBER 14,1973
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C* LATEST REVISION
C*

C* SEPTEMBER 14, 1973 *

C* *

C*****

C

C

C

C*****

C

C

C*-*-

C

C

C*****

C

C

C

000002 COMMON / GUSTA / TDOT8.DOT8
000002 DIMENSION AA(151),BB(151),V(151),W(151)

C *****

C

C

C DEFINE THE GUST VELOCITY PROFILE

C

000002 READ(5,7) ALPHA,NU,NRB
000014 7 FORMAT(5X,F5.1,2I5)
000014 PI=3.1416
000015 PIA=PI/ALPHA
000017 RR=64.0
000020 ZNU=FLOAT(NU)
000021 ZNUA=-(ZNU+ALPHA)
000023 VM=2.0578
000025 VM2=0.5*VM
000026 J=NRB+1
000031 WRITE(6,10)
000034 10 FORMAT(1H1,30X,21HGUST VELOCITY PROFILE.//)

C

C DEFINE CENTERLINE VELOCITY(CORRESPONDS TO X=0)

C

000034 AA(1)=VM2*(1.0-COS(PIA*ZNUA))
000042 BB(1)=AA(1)
000044 V(1)=AA(1)+BB(1)
000046 WRITE(6,20) AA(1),BB(1),V(1)
000057 20 FORMAT(5X,1H1,23X,2HAA,34X,2HBB,36X,1HV//4X,3H 1,3(12X,E24.15))

C
C DEFINE THE OFF-CENTERLINE VELOCITIES
C

000057 DO 40 I=2,J
000061 ZN=FLOAT(I-1)
000063 AA(I)=VM2*(1.0-COS(PIA*(-ZN+ZNUA)))
000074 BB(I)=VM2*(1.0-COS(PIA*(ZN+ZNUA)))
000104 V(I)=AA(I)+BB(I)
000106 WRITE(6,30) I,AA(I),BB(I),V(I)
000122 30 FORMAT(4X,I3,3(12X,E24.15))
000122 40 CONTINUE

C
C COMPUTE THE ANGULAR VELOCITIES
C

000125 WRITE(6,50)
000130 50 FORMAT(1H1,30X,23HGUST ROTATIONAL IMPULSE.//)
000130 W(1)=1.5/(RB*RB)*(V(2)-V(1))
000134 WRITE(6,60) W(1)
000142 60 FORMAT(2X,1HW//4X,3H 1,12X,E24.15)
000142 DO 80 K=2,NRB
000144 ZN=FLOAT(K)
000145 W(K)=W(K-1)+1.5/(RB*RB)*ZN*(V(K+1)-V(K))
000155 WRITE(6,70) K,W(K)
000165 70 FORMAT(4X,I3,12X,E24.15)
000165 80 CONTINUE

C
000170 DDIB=0.01*W(150)
000172 DDIB=0.01*W(150)

C
000173 RETURN
000174 END

SUBROUTINE DIFSUB(N,T,H,Y,WK,JSTART,HMIN,HMAX,EPS,MTH,MAXDER,IERR)

```
C*****00000020
C# 00000030
C# PURPOSE 00000040
C# 00000050
C# SUBROUTINE DIFSUB INTEGRATES A SYSTEM OF N FIRST ORDER ORDINARY 00000060
C# DIFFERENTIAL EQUATIONS OVER ONE STEP OF LENGTH H AT EACH CALL. 00000070
C# H CAN BE SPECIFIED BY THE USER FOR ANY OR ALL STEPS, BUT, IT MAY 00000080
C# BE INCREASED OR DECREASED BY DIFSUB WITHIN THE RANGE HMIN TO HMAX. 00000090
C# THE STEP ADJUSTMENT IS USED TO ACHIEVE AS LARGE A STEP AS 00000100
C# POSSIBLE, WHILE NOT COMMITTING A SINGLE STEP ERROR WHICH IS 00000110
C# LARGER THAN EPS IN THE L-2 NORM. THE USER SPECIFIED PARAMETER MTH 00000120
C# INDICATES WHETHER THE SYSTEM IS TO BE SOLVED BY NON-STIFF OR 00000130
C# STIFF METHODS. 00000140
C# 00000150
C# PROPRIETARY 00000160
C# 00000170
C# NONE. 00000180
C# 00000190
C# USE 00000200
C# 00000210
C# CALL DIFSUB(N,T,H,Y,WK,JSTART,HMIN,HMAX,EPS,MTH,MAXDER,IERR). 00000220
C# 00000230
C# PARAMETERS 00000240
C# 00000250
C# N-----THE NUMBER OF FIRST ORDER DIFFERENTIAL EQUATIONS. 00000260
C# 00000270
C# T-----THE INDEPENDENT VARIABLE. 00000280
C# 00000290
C# H-----ON INPUT, THE STEP LENGTH TO BE ATTEMPTED ON THIS STEP. 00000300
C# ON OUTPUT, THE STEP LENGTH DIFSUB RECOMMENDS FOR THE NEXT 00000310
C# STEP. DIFSUB WILL USE ITS RECOMMENDED STEP LENGTH UNLESS 00000320
C# THE USER MODIFIES IT. 00000330
C# 00000340
C# H MAY BE ADJUSTED UP AS FAR AS HMAX OR DOWN AS FAR AS HMIN 00000350
C# BY DIFSUB IN ORDER TO ACHIEVE AN ECONOMICAL INTEGRATION. 00000360
C# HOWEVER, IF THE USER SPECIFIES AN H VALUE, IT WILL BE USED 00000370
C# PROVIDED SUCH AN H DOES NOT CAUSE A SINGLE STEP ERROR 00000380
C# GREATER THAN EPS. THE USER IS REQUIRED TO SUPPLY H ONLY ON 00000390
C# THE FIRST CALL. TO SAVE COMPUTER TIME, THE USER IS ADVISED 00000400
C# TO USE A SMALL STEP (SAY .125E-06) FOR THE FIRST CALL. 00000410
C# AFTER THE FIRST CALL THE STEP LENGTH WILL BE INCREASED BY 00000420
```

```

C*      DIFSUB. *00000430
C*      *00000440
C*      Y-----AN ARRAY OF 8*N LOCATIONS CONTAINING THE DEPENDENT *00000450
C*      VARIABLES AND THEIR SCALED DERIVATIVES. Y(N*(J-1)+1) *00000460
C*      CONTAINS J-TH DERIVATIVE OF Y(I) SCALED BY *00000470
C*      H**J/FACTORIAL(J), WHERE H IS THE CURRENT STEP SIZE. ONLY *00000480
C*      Y(I), I=1,N MUST BE PROVIDED BY THE CALLING PROGRAM ON THE *00000490
C*      FIRST CALL (INITIAL VALUES). DERIVATIVES ARE CALCULATED BY *00000500
C*      DIFSUB BY NUMERICAL DIFFERENCING AND DO NOT NEED TO BE *00000510
C*      PROVIDED BY THE USER. IF IT IS DESIRED TO INTERPOLATE TO *00000520
C*      NON-MESH POINTS, THESE VALUES CAN BE USED. IF THE CURRENT *00000530
C*      STEP SIZE IS H AND THE VALUE AT T+E IS NEEDED, FORM S=E/H *00000540
C*      AND COMPUTE *00000550
C*      *00000560
C*      NO *00000570
C*      Y(I)(T+E)=SUM Y(N*(J-1)+I)*S**J *00000580
C*      J=0 *00000590
C*      *00000600
C*      WHENEVER DIFSUB IS CALLED WITH JSTART=0, Y(I), I=N+1,8*N *00000610
C*      IS SET TO 0. *00000620
C*      *00000630
C*      WK-----A WORK STORAGE ARRAY OF DIMENSION N*(N+14). THE USER NEED *00000640
C*      ONLY DIMENSION THE ARRAY CORRECTLY. THE ARRAY IS ZERO *00000650
C*      FILLED OR INITIALIZED AND USED BY DIFSUB. THE USER SHOULD *00000660
C*      NOT PUT ANYTHING INTO THE ARRAY. THE ARRAY IS USED TO *00000670
C*      HOLD THE FOLLOWING *00000680
C*      *00000690
C*      WK(I)-WK(8*N) *00000700
C*      THE Y ARRAY VALUES ARE SAVED IN THE EVENT THAT A STEP *00000710
C*      MUST BE REPEATED. *00000720
C*      *00000730
C*      WK(8*N+1)-WK(9*N) *00000740
C*      HOLDS CORRECTION TERMS IN THE CORRECTOR LOOP. *00000750
C*      *00000760
C*      WK(9*N+1)-WK(10*N) *00000770
C*      HOLDS SUMS OF ALL CORRECTION TERMS IN THE PREVIOUS *00000780
C*      STEP AFTER THE SUMS HAVE BEEN ACCUMULATED IN *00000790
C*      WK(13*N+1), I=1,N FOR THE CURRENT STEP. THIS ENABLES *00000800
C*      THE BACKWARDS DIFFERENCE OF WK(13*N+1) TO BE FORMED *00000810
C*      AND USED TO ESTIMATE THE STEP SIZE FOR ONE ORDER *00000820
C*      HIGHER THAN CURRENT. *00000830
C*      *00000840
C*      WK(10*N+1)-WK(11*N) *00000850

```

```

C*          HOLDS THE DERIVATIVES OF Y(I) WITH RESPECT TO T WHEN #00000860
C*          THEY ARE COMPUTED BY DIFFUN. #00000870
C* #00000880
C*          WK(11*N+1)-WK(12*N) #00000890
C*          HOLDS THE DERIVATIVES OF Y(I) WITH RESPECT TO T #00000900
C*          DURING JACOBIAN EVALUATION. #00000910
C* #00000920
C*          WK(12*N+1)-WK(13*N) #00000930
C*          HOLDS THE MAXIMUM ABSOLUTE VALUE OF Y(I) UP TO AND #00000940
C*          INCLUDING THE PRESENT STEP. #00000950
C* #00000960
C*          WK(13*N+1)-WK(14*N) #00000970
C*          HOLDS THE SINGLE STEP ERROR ESTIMATES #00000980
C*          WK(13*N+1)=Y(N*(K-1)+1)*H**K/IFACTORIAL(K-1)*A(K) #00000990
C*          WHERE K IS THE HIGHEST DERIVATIVE AVAILABLE. #00001000
C* #00001010
C*          WK(14*N+1)-WK(14*N*N**2) #00001020
C*          USED FOR STIFF METHODS(MTH=1,2) TO HOLD JACOBIAN BY #00001030
C*          COLUMNS. WK(N*(13+J)+1)={D/DY(J)}DY(I) .I=1..N J=1..N #00001040
C* #00001050
C*          JSTART=AN INPUT/OUTPUT CODE USED AS FOLLOWS #00001060
C* #00001070
C*          INPUT #00001080
C*          #00001090
C*          =-1 REPEAT THE LAST STEP WITH A NEW H. #00001100
C* #00001110
C*          =0 PERFORM THE FIRST STEP. THE FIRST CALL MUST BE DONE #00001120
C*          WITH THIS VALUE TO PERMIT PROPER INITIALIZATION. #00001130
C* #00001140
C*          =+1 TAKE A NEW STEP CONTINUING FROM THE LAST. #00001150
C* #00001160
C*          OUTPUT #00001170
C* #00001180
C*          =NQ THE CURRENT ORDER OF THE INTEGRATION METHOD ON RETURN #00001190
C*          ALSO, THE MAXIMUM DERIVATIVE AVAILABLE. #00001200
C* #00001210
C*          HMIN---THE MINIMUM STEP SIZE THAT WILL BE USED FOR THE #00001220
C*          INTEGRATION. NOTE THAT ON STARTING HMIN MUST BE MUCH #00001230
C*          SMALLER (SAY .125E-09) THAN THE AVERAGE H EXPECTED SINCE #00001240
C*          A FIRST ORDER METHOD IS USED INITIALLY. #00001250
C* #00001260
C*          HMAX---THE MAXIMUM STEP SIZE THE USER WILL ALLOW. #00001270
C* #00001280

```

C*	EPS----	THE ERROR TEST CONSTANT. SINGLE STEP ERROR ESTIMATES	*00001290
C*		WK(N*13+1) DIVIDED BY WK(N*12+1) MUST BE LESS THAN EPS IN	*00001300
C*		THE EUCLIDEAN NORM. STEP AND/OR ORDER IS ADJUSTED TO	*00001310
C*		ACHIEVE THIS ERROR BOUND.	*00001320
C*			*00001330
C*	MTH----	THE METHOD INDICATOR.	*00001340
C*			*00001350
C*	=0	AN ADAMS PREDICTOR-CORRECTOR FOR NON-STIFF SYSTEMS.	*00001360
C*			*00001370
C*	=1	A MULTI-STEP METHOD FOR STIFF SYSTEMS. IT WILL ALSO	*00001380
C*		WORK FOR NON-STIFF SYSTEMS. A USER SUPPLIED ROUTINE	*00001390
C*		PEDERV IS REQUIRED TO CALCULATE THE PARTIAL	*00001400
C*		DERIVATIVES (D/DY(J))DY(I), THAT IS, THE JACOBIAN.	*00001410
C*		ANOTHER USER SUPPLIED ROUTINE MATINV IS REQUIRED TO	*00001420
C*		INVERT THE JACOBIAN.	*00001430
C*			*00001440
C*	=2	THE SAME AS MTH=1 EXCEPT THAT THE PARTIAL DERIVATIVES	*00001450
C*		ARE CALCULATED IN DIFSUB BY NUMERICAL DIFFERENCING	*00001460
C*		AND HENCE PEDERV IS NOT CALLED. HOWEVER A DUMMY	*00001470
C*		SUBROUTINE NAMED PEDERV SHOULD BE SUPPLIED. MATINV IS	*00001480
C*		STILL NECESSARY TO INVERT THE JACOBIAN.	*00001490
C*			*00001500
C*	MAXDER-	THE MAXIMUM ORDER INTEGRATION THE USER WISHES TO BE USED.	*00001510
C*		FOR NON-STIFF METHODS(MTH=0) IT MUST SATISFY	*00001520
C*		1.GE.MAXDER.LE.7 AND FOR STIFF METHODS(MTH=1,2) IT MUST	*00001530
C*		SATISFY 1.GE.MAXDER.LE.6.	*00001540
C*			*00001550
C*	IERR----	A COMPLETION CODE WITH THE FOLLOWING MEANINGS	*00001560
C*			*00001570
C*	=+1	THE STEP WAS SUCCESSFUL. ALL REQUIREMENTS WERE MET.	*00001580
C*			*00001590
C*	=-1	THE STEP WAS TAKEN WITH H=HMIN; BUT, THE REQUESTED	*00001600
C*		ERROR, EPS, WAS NOT ACHIEVED.	*00001610
C*			*00001620
C*	=-2	THE MAXIMUM ORDER SPECIFIED, MAXDER, WAS TOO LARGE.	*00001630
C*		OBSERVE THE CONSTRAINTS IN THE DEFINITION OF MAXDER.	*00001640
C*			*00001650
C*	=-3	CORRECTOR CONVERGENCE COULD NOT BE ACHIEVED FOR	*00001660
C*		H.GT.HMIN.	*00001670
C*			*00001680
C*	=-4	THE REQUESTED ERROR IS SMALLER THAN CAN BE HANDLED	*00001690
C*		FOR THE CONSTRAINTS PLACED ON THE PARAMETERS. EXAMINE	*00001700
C*		THE PARAMETERS TO DETERMINE IF SOME MAY BE RELAXED	*00001710


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C*          AND TRY AGAIN.
C*
C*  NOTE: THE FOLLOWING POINTS SHOULD BE KEPT IN MIND WHEN CHOOSING
C*        THE PARAMETERS.
C*
C*        HIGHER ORDER INTEGRATIONS (MAXDER) AND SMALL STEP SIZES
C*        (H) REQUIRE MORE CALCULATIONS AND HENCE MORE TIME, BUT
C*        THEY RESULT IN BETTER ACCURACY.
C*
C*        SMALL ERROR TOLERANCES (EPS) REQUIRE HIGHER ORDER
C*        INTEGRATIONS AND/OR SMALLER STEP SIZE.
C*
C*        THE USER SHOULD NOT ATTEMPT TO CHOOSE THE BEST STEP SIZE
C*        OR RESTRICT THE ORDER TO ATTEMPT AN OPTIMAL COMBINATION.
C*        AS MUCH AS POSSIBLE, SUCH DECISIONS SHOULD BE LEFT TO
C*        DIFSUB. DIFSUB BASES ITS DECISIONS ON PRECISE CALCULATIONS
C*        AND OPTIMIZES STEP SIZE AND ORDER TO ACHIEVE THE ERROR
C*        TOLERANCE WITH ECONOMICAL CALCULATIONS.
C*
C*        IF NECESSITY MAY REQUIRE SOME PARAMETERS TO BE
C*        CONSTRAINED; THOSE NOT CONSTRAINED SHOULD BE FREE TO VARY
C*        AS MUCH AS POSSIBLE TO PERMIT DIFSUB TO OPTIMIZE THEM.
C*
C*  REQUIRED ROUTINES
C*
C*  ALL OF THE FOLLOWING REQUIRED ROUTINES MUST BE SUPPLIED BY THE
C*  USER EVEN IF ONLY AS DUMMY SUBROUTINES.
C*
C*  SUBROUTINE DIFFUN(T,Y,DY)
C*
C*      T--INDEPENDENT VARIABLE
C*
C*      Y--ARRAY OF DEPENDENT VARIABLES AND SCALED DERIVATIVES.
C*      THIS ARRAY IS THE SAME AS THE Y ARRAY FOR DIFSUB AND IS
C*      DIMENSIONED THE SAME.
C*
C*      DY--ARRAY OF DIMENSION N WHICH HOLDS THE DERIVATIVES OF
C*      Y(I), I=1..N WITH RESPECT TO T.
C*
C*      DIFFUN MUST ALWAYS BE SUPPLIED. IT COMPUTES THE DERIVATIVE
C*      OF Y(I) WITH RESPECT TO T. RETURNS IT IN DY(I) TO DIFSUB.
C*
C*  SUBROUTINE MATINV(PW,N,J)

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*00001720
*00001730
*00001740
*00001750
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*00001980
*00001990
*00002000
*00002010
*00002020
*00002030
*00002040
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*00002060
*00002070
*00002080
*00002090
*00002100
*00002110
*00002120
*00002130
*00002140

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C*          PW--ARRAY CONTAINING THE JACOBIAN PASSED FROM DIFSUB.      *00002150
C*          DIMENSIONED PW(N,N)                                       *00002160
C*          N--THE SIZE OF THE SYSTEM.                                  *00002170
C*          J--RETURN CODE                                             *00002180
C*          =1 IF INVERSION OF PW SUCCESSFUL (DET(PW).NE.0).          *00002190
C*          =-1 IF INVERSION OF PW UNSUCCESSFUL (DET(PW).EQ.0).      *00002200
C*          MATINV MUST BE PROVIDED FOR STIFF METHODS(MTH=1,2) WHILE *00002210
C*          ONLY A DUMMY SUBROUTINE IS NECESSARY FOR NON-STIFF METHODS *00002220
C*          (MTH=0). MATINV INVERTS THE JACOBIAN AND RETURNS THE      *00002230
C*          INVERSE IN PW TO DIFSUB.                                    *00002240
C*          SUBROUTINE PEDERV(T,Y,PW,N)                                  *00002250
C*          T--INDEPENDENT VARIABLE.                                    *00002260
C*          Y--ARRAY OF DEPENDENT VARIABLES AND FIRST SEVEN          *00002270
C*          DERIVATIVES. THIS ARRAY IS THE SAME AS THE Y ARRAY FOR   *00002280
C*          DIFSUB AND IS DIMENSIONED THE SAME.                       *00002290
C*          PW--ARRAY DIMENSIONED N**2 CONTAINING THE JACOBIAN BY    *00002300
C*          COLUMNS. PW(N**2).                                       *00002310
C*          N--THE SIZE OF THE SYSTEM.                                  *00002320
C*          PEDERV MUST BE PROVIDED FOR STIFF METHODS(MTH=1,2) WHILE *00002330
C*          ONLY A DUMMY SUBROUTINE IS REQUIRED FOR NON-STIFF METHODS *00002340
C*          (MTH=0). PEDERV COMPUTES THE PARTIAL DERIVATIVES OF DY(I) *00002350
C*          I=1,N AND STORES THEM BY COLUMNS IN THE PW ARRAY.       *00002360
C*          PW(N*(J-1)+I)={D/DY(I)}DY(I) I=1,N , J=1,N              *00002370
C*          AUTHOR/IMPLEMENTER                                         *00002380
C*          S.BAUDENDISTEL/G.W.HAIGLER                                 *00002390
C*          LANGUAGE                                                    *00002400
C*          FORTRAN                                                    *00002410
C*          *00002420
C*          *00002430
C*          *00002440
C*          *00002450
C*          *00002460
C*          *00002470
C*          *00002480
C*          *00002490
C*          *00002500
C*          *00002510
C*          *00002520
C*          *00002530
C*          *00002540
C*          *00002550
C*          *00002560
C*          *00002570

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C*                                     *00002580
C* DATE RELEASED                       *00002590
C*                                     *00002600
C*   MAY 25, 1973                       *00002610
C*                                     *00002620
C* LATEST REVISION                      *00002630
C*                                     *00002640
C*   MAY 25, 1973                       *00002650
C*                                     *00002660
C*****                                *00002670
C                                     *00002680
00017. DIMENSION Y(1),WK(1),A(8),PERTST(42) *00002690
C                                     *00002700
C*****                                *00002710
C* THE COEFFICIENTS IN PERTST ARE USED IN SELECTING THE STEP AND *00002720
C* ORDER, THEREFORE ONLY ABOUT ONE PERCENT ACCURACY IS NEEDED. *00002730
C*****                                *00002740
C*                                     *00002750
C*****                                *00002760
C* THIS ROW IS USED TO CALCULATE E FOR STIFF SYSTEMS *00002770
C*                                     *00002780
00017 DATA PERTST /2.0,4.5,7.333,10.42,13.7,17,15,1.0. *00002790
C*****                                *00002800
C* THIS ROW IS USED TO CALCULATE E FOR NON-STIFF SYSTEMS *00002810
C*                                     *00002820
C*   1      2.0,12.0,24.0,37.89,53.33,70.08,87.97, *00002830
C*****                                *00002840
C* THIS ROW IS USED TO CALCULATE EUP FOR STIFF SYSTEMS *00002850
C*                                     *00002860
C*   2      3.0,6.0,9.167,12.5,15.98,1.0,1.0. *00002870
C*****                                *00002880
C* THIS ROW IS USED TO CALCULATE EUP FOR NON-STIFF SYSTEMS *00002890
C*                                     *00002900
C*   3      12.0,24.0,37.89,53.33,70.08,87.97,1.0, *00002910
C*****                                *00002920
C* THIS ROW IS USED TO CALCULATE EDWN FOR STIFF SYSTEMS *00002930
C*                                     *00002940
C*   4      1.,1.,0.5,0.1667,0.04133,0.008267,1.0, *00002950
C*****                                *00002960
C* THIS ROW IS USED TO CALCULATE EDWN FOR NON-STIFF SYSTEMS *00002970
C*                                     *00002980
C*   5      1.0,1.0,2.0,1.0,.3157,.07407,.0139/ *00002990
C*****                                *00003000

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000017 DATA A(2) / -1.0 / 00003010
000017 IRET=1 00003020
000017 IERR=+1; 00003030
000020 IF(IJSTART.LE.0)GOTO 25 00003040
C***** 00003050
C* BEGIN BY SAVING INFORMATION FOR POSSIBLE RESTARTS AND CHANGING *00003060
C* H BY THE FACTOR R IF THE CALLER HAS CHANGED H. ALL VARIABLES *00003070
C* DEPENDENT ON H MUST ALSO BE CHANGED. *00003080
C* HNFH IS THE STEP SIZE THAT WAS USED ON THE LAST CALL. *00003090
C***** 00003100
000022 5 NN=N*K 00003110
000025 DO 10 I=1,NN 00003120
000026 10 WK(I)=Y(I) 00003130
000033 HOLD=HNFH 00003140
000034 IF(H.EQ.HOLD)GOTO 20 00003150
000036 15 RACUM=H/HOLD 00003160
000037 IRET1=1 00003170
000041 GOTO 415 00003180
000041 20 NOLD=N 00003190
000042 TOL=1 00003200
000043 RALIM=1. 00003210
000045 IF(IJSTART.GT.0)GOTO 170 00003220
000047 GOTO 55 00003230
000047 25 IF(IJSTART.EQ.-1)GOTO 50 00003240
C***** 00003250
C* ON THE FIRST AND ANY SUBSEQUENT CALL WITH JSTART=0, THE ORDER IS *00003260
C* SET TO 1 AND THE INITIAL DERIVATIVES ARE CALCULATED. ARRAY ACCESS *00003270
C* CONSTANTS ARE CALCULATED AND ARRAYS Y,WK ARE ZEROED OR INITIALIZED *00003280
C***** 00003290
000051 NQ=1 00003300
000051 NR=N*B 00003310
000053 N9=N8+N 00003320
000054 N10=N9+N 00003330
000055 N11=N10+N 00003340
000056 N12=N11+N 00003350
000057 N13=N12+N 00003360
000060 N14=N13+N 00003370
000061 N10P1=N10+1 00003380
000063 N11P1=N11+1 00003390
000065 N14P1=N14+1 00003400
000066 NN=N*(N+14) 00003410
000071 DO 30 I=1,NN 00003420
000073 30 WK(I)=0. 00003430
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000077	NN=N+1	00000360
000100	DO 35 I=NN,N8	00000370
000101	35 Y(I)=0	00000380
000105	DO 40 I=1,N	00000390
000106	40 WK(N12+I)=AMAX1(1.,ABS(Y(I)))	00000400
000131	CALL DIFFUN(T,Y,WK(N10P1))	00000410
000125	DO 45 I=1,N	00000420
000132	45 Y(N+I)=WK(N10+I)*H	00000430
000141	HNEW=H	00000440
000141	K=2	00000450
000143	GOTO 5	00000460
	C*****	00000470
	C* REPEAT LAST STEP BY RESTORING SAVED INFORMATION.	*00000480
	C*****	00000490
000143	50 IF(NQ.EQ.NQOLD)JSTART=1	00000500
000146	T=TOLO	00000510
000147	NQ=NQOLD	00000520
000150	K=NQ+1	00000530
000152	GOTO 15	00000540
	C*****	00000550
	C* SET THE COEFFICIENTS THAT DETERMINE THE ORDER AND THE METHOD	*00000560
	C* TYPE. CHECK FOR EXCESSIVE ORDER. THE LAST TWO STATEMENTS OF	*00000570
	C* THIS SECTION SET IWEVAL .GT.0 IF THE JACOBIAN IS TO BE RE-	*00000580
	C* EVALUATED BECAUSE OF THE ORDER CHANGE. AND THEN REPEAT THE	*00000590
	C* INTEGRATION STEP IF IT HAS NOT YET BEEN DONE (IRET=1) OR SKIP TO A	*00000600
	C* FINAL SCALING BEFORE EXIT IF IT HAS BEEN COMPLETED (IRET=2).	*00000610
	C*****	00000620
000153	55 IF(MTH.EQ.0)GOTO 70	00000630
000155	IF(NQ-2)130,135,60	00000640
000157	60 IF(NQ-4)140,145,65	00000650
000162	65 IF(NQ-6)150,155,90	00000660
000165	70 IF(NQ-2)95,100,75	00000670
000170	75 IF(NQ-4)105,110,80	00000680
000173	80 IF(NQ-6)115,120,85	00000690
000176	85 IF(NQ.EQ.7)GOTO 125	00000700
000200	90 IERR=-2	00000710
000202	RETURN	00000720
	C*****	00000730
	C* THE FOLLOWING COEFFICIENTS SHOULD BE DEFINED TO THE MAXIMUM	*00000740
	C* ACCURACY PERMITTED BY THE MACHINE. THEY ARE. IN THE ORDER USED..	*00000750
	C*	*00000760
	C* THE FOLLOWING ARE FOR NON-STIFF METHODS	*00000770
	C*	*00000780

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C* -1	*00000790
C* -1/2.-1/2	*00000800
C* -5/12,-3/4,-1/6	*00000810
C* -3/8,-11/12,-1/3,-1/24	*00000820
C* -251/720,-25/4,-35/72,-5/48,-1/120	*00000830
C* -95/288,-137/120,-5/8,-17/96,-1/40,-1/720	*00000840
C* -19087/63480,-49/40,-203/270,-49/192,-7/144,-7/1440,-1/5040	*00000850
C*	*00000860
C* THE FOLLOWING ARE FOR STIFF METHODS	*00000870
C*	*00000880
C* -1	*00000890
C* -2/3,-1/3	*00000900
C* -6/11,-6/11,-1/11	*00000910
C* -12/25,-7/10,-1/5,-1/50	*00000920
C* -120/274,-225/274,-85/274,-15/274,-1/274	*00000930
C* -180/441,-58/63,-15/36,-25/252,-3/252,-1/1764	*00000940
C*****	*00000950
000202 95 A(1) = -1.0	00000960
000204 GOTO 160	00000970
000206 100 A(1) = -0.500000000	00000980
000208 A(3) = -0.500000000	00000990
000210 GOTO 160	00010000
000212 105 A(1) = -0.416666666666667	00010100
000214 A(3) = -0.750000000	00010200
000216 A(4) = -0.166666666666667	00010300
000218 GOTO 160	00010400
000220 110 A(1) = -0.375000000	00010500
000222 A(3) = -0.916666666666667	00010600
000224 A(4) = -0.333333333333333	00010700
000226 A(5) = -0.041666666666667	00010800
000228 GOTO 160	00010900
000230 115 A(1) = -0.348611111111111	00011000
000232 A(3) = -1.041666666666667	00011100
000234 A(4) = -0.486111111111111	00011200
000236 A(5) = -0.104166666666667	00011300
000238 A(6) = -0.00833333333333333	00011400
000240 GOTO 160	00011500
000242 120 A(1) = -0.329861111111111	00011600
000244 A(3) = -1.141666666666667	00011700
000246 A(4) = -0.625000000	00011800
000248 A(5) = -0.177083333333333	00011900
000250 A(6) = -0.025000000	00012000
000252 A(7) = -0.00138888888888889	00012100

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000244      GOTO 160
000245  125 A(1) = -0.3155919312169312
000246      A(3) = -1.235000000
000250      A(4) = -0.7518518518518519
000251      A(5) = -0.2552083333333333
000253      A(6) = -0.0486111111111111
000254      A(7) = -0.0048611111111111
000256      A(8) = -0.0001984126984126984
000260      GOTO 160
000260  130 A(1) = -1.000000000
000262      GOTO 160
000262  135 A(1) = -0.6666666666666667
000263      A(3) = -0.3333333333333333
000265      GOTO 160
000266  140 A(1) = -0.5454545454545455
000267      A(3) = A(1)
000271      A(4) = -0.09090909090909091
000273      GOTO 160
000273  145 A(1) = -0.480000000
000274      A(3) = -0.700000000
000276      A(4) = -0.200000000
000277      A(5) = -0.020000000
000301      GOTO 160
000302  150 A(1) = -0.437955204379552
000303      A(3) = -0.8211678832116788
000305      A(4) = -0.3102189781021898
000306      A(5) = -0.05474452554744526
000310      A(6) = -0.0036496350364963504
000312      GOTO 160
000312  155 A(1) = -0.4081632653061225
000313      A(3) = -0.9206349206349206
000315      A(4) = -0.4166666666666667
000316      A(5) = -0.092063492063492
000320      A(6) = -0.0119047619047619
000321      A(7) = -0.000566893424036282
000323  160 K=NO+1
000325      I0008=K
*****
C* MTYP=1 FOR STIFF METHODS AND MTYP=2 FOR NON-STIFF METHODS
*****
000326      MTYP=1
000327      IF(MTH.EQ.0)MTYP=2
000332      ED=NO

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000333      ENQ1=.5/EDWN      00001650
000335      ENQ2=.5/(EDWN+1.) 00001660
000341      ENQ3=.5/(EDWN+2.) 00001670
000344      L=7*(MTYP-1)+NQ 00001680
C*****00001690
C* E IS A COMPARISON FOR ERRORS OF THE CURRENT ORDER NQ. EUP IS *00001700
C* TO TEST FOR INCREASING THE ORDER, EDWN FOR DECREASING THE ORDER. *00001710
C*****00001720
000347      E=(PERTST(L)*EPS)**2 00001730
000353      EUP=(PERTST(L+1)*EPS)**2 00001740
000355      EDWN=(PERTST(L+2)*EPS)**2 00001750
000357      IF(EDWN.EQ.0.)GOTO 430 00001760
000360      BND=EPS*ENQ3/N 00001770
000363      165 THEVAL=MTH 00001780
000364      IF(IRET.GT.1)GOTO 380 00001790
C*****00001800
C* THIS SECTION COMPUTES THE PREDICTED VALUES BY EFFECTIVELY *00001810
C* MULTIPLYING THE SAVED INFORMATION BY THE PASCAL TRIANGLE MATRIX *00001820
C*****00001830
000370      173 T=T+H 00001840
000371      L=N*(K-1) 00001850
000375      DO 175 J=2,K 00001860
000376      NN=L 00001870
000400      DO 175 J1=J,K 00001880
000401      NN=NN-N 00001890
000403      DO 175 I=1,N 00001900
000404      NT=NN+I 00001910
000405      175 Y(NT)=Y(NT)+Y(NT+N) 00001920
C*****00001930
C* UP TO 3 CORRECTOR ITERATIONS ARE TAKEN. CONVERGENCE IS TESTED BY *00001940
C* REQUIRING CHANGES TO BE LESS THAN BND WHICH IS DEPENDENT ON THE *00001950
C* ERROR TEST CONSTANT. THE SUM OF THE CORRECTIONS IS ACCUMULATED IN *00001960
C* THE ARRAY WK(N13+I). IT IS EQUAL TO THE K-TH DERIVATIVE OF Y *00001970
C* MULTIPLIED BY H**K/(FACTORIAL(K-1)*A(K)), AND IS PROPORTIONAL TO *00001980
C* THE ACTUAL ERRORS TO THE LOWEST POWER OF H PRESENT. (H**K) *00001990
C*****00002000
000420      DO 180 I=1,N 00002010
000422      180 WK(N13+I)=0. 00002020
000427      DO 260 L=1,3 00002030
000430      CALL DIFFUN(T,Y,WK(N10+I)) 00002040
C*****00002050
C* IF THERE HAS BEEN A CHANGE OF ORDER OR THERE HAS BEEN TROUBLE *00002060
C* WITH CONVERGENCE, THE JACOBIAN IS RE-EVALUATED PRIOR TO STARTING. *00002070

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C* THE CORRECTOR ITERATION IN THE CASE OF STIFF METHODS. IWEVAL IS *00002080
C* THEN SET TO -1 AS AN INDICATOR THAT IT HAS BEEN DONE. *00002090
C*****00002100
000434 IF(IWEVAL.LT.1)GOTO 220 00002110
000442 IF(MTH.EQ.2)GOTO 200 00002120
000445 CALL PEDERV(T,Y,WK(N14P1),N) 00002130
000452 R=A(I)*H 00002140
000457 NN=N*N 00002150
000461 DO 185 I=1,NN 00002160
000463 185 WK(N14+I)=WK(N14+I)*R 00002170
C*****00002180
C* ADD THE IDENTITY MATRIX TO THE JACOBIAN AND INVERT *00002190
C*****00002200
000471 190 NN=N13 00002210
000472 J1=N+1 00002220
000474 DO 195 I=1,N 00002230
000476 NN=NN+J1 00002240
000500 195 WK(NN)=1.+WK(NN) 00002250
000505 IWEVAL=-1 00002260
000506 CALL MAT(INV(WK(N14P1),N,J1) 00002270
000514 IF(J1.GT.0)GOTO 220 00002280
000522 GOTO 265 00002290
C*****00002300
C* EVALUATE THE JACOBIAN BY NUMERICAL DIFFERENCING. R IS THE CHANGE *00002310
C* MADE TO THE ELEMENT OF Y. IT IS EPS RELATIVE TO Y WITH A MINIMUM OF *00002320
C* EPS**2 *00002330
C*****00002340
000522 200 DO 205 I=1,N 00002350
000524 205 WK(NB+I)=Y(I) 00002360
000532 J1=N13 00002370
000533 DO 215 J=1,N 00002380
000535 R=EPS*AMAX1(EPS,ABS(WK(NB+J))) 00002390
000544 Y(J)=Y(J)+R 00002400
000547 D=A(I)*H/R 00002410
000551 CALL DIFFUN(T,Y,WK(N11P1)) 00002420
000556 J1=J1+N 00002430
000563 DO 210 I=1,N 00002440
000565 NN=J1+I 00002450
000566 210 WK(NN)=(WK(N11+I)-WK(N10+I))*D 00002460
000602 215 Y(J)=WK(NB+J) 00002470
000610 GOTO 190 00002480
000610 220 IF(MTH.NE.0)GOTO 230 00002490
000612 DO 225 I=1,N 00002500

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000613 225 WK(NB+I)=Y(N+I)-WK(N10+I)*H 00002510
000625 GOTO 250 00002520
000625 230 DO 235 I=1,N 00002530
000627 235 WK(N11+I)=Y(N+I)-WK(N10+I)*H 00002540
000641 DO 245 I=1,N 00002550
000642 D=0. 00002560
000643 NN=N13+I 00002570
000645 DO 240 J=1,N 00002580
000646 NN=NN+N 00002590
000647 240 D=D+WK(NN)*WK(N11+J) 00002600
000657 245 WK(NB+I)=D 00002610
000664 25) NT=N 00002620
C*****00002630
C* CORRECT AND SEE IF ALL CHANGES ARE LESS THAN BND RELATIVE TO *00002640
C* WK(N12+I). IF SO, THE CORRECTOR IS SAID TO HAVE CONVERGED. *00002650
C*****00002660
000665 DO 255 I=1,N 00002670
000667 Y(I)=Y(I)+A(I)*WK(NB+I) 00002680
000674 Y(N+I)=Y(N+I)-WK(NB+I) 00002690
000701 WK(N13+I)=WK(N13+I)+WK(NB+I) 00002700
000705 IF (ABS(WK(NB+I)).LE.BND*WK(N12+I)) NT=NT-1 00002710
000717 255 CONTINUE 00002720
000722 IF (NT.LE.0) GOTO 285 00002730
000723 260 CONTINUE 00002740
C*****00002750
C* THE CORRECTOR ITERATION FAILED TO CONVERGE IN 3 TRIES. VARIOUS *00002760
C* POSSIBILITIES ARE CHECKED FOR. IF H IS ALREADY HMIN AND THIS IS *00002770
C* EITHER ADAMS METHOD OR THE STIFF METHOD IN WHICH THE JACOBIAN HAS *00002780
C* ALREADY BEEN RE-EVALUATED, A NO CONVERGENCE EXIT IS TAKEN. *00002790
C* OTHERWISE THE JACOBIAN IS RE-EVALUATED AND/OR THE STEP IS REDUCED *00002800
C* TO TRY AND GET CONVERGENCE. *00002810
C*****00002820
000725 265 T=TOLD 00002830
000726 IF (H.LE.HMIN*1.00001.AND.(IWEVAL-MTYP).LT.-1) GOTO 270 00002840
000743 IF (N1H.EQ.0.OR.IWEVAL.NE.0) RACUM=RACUM*.25 00002850
000752 IWEVAL=MTH 00002860
000753 IRET=2 00002870
000755 GOTO 415 00002880
000755 270 IERR=-3 00002890
000757 275 NN=N*K 00002900
000762 DO 280 I=1,NN 00002910
000763 280 Y(I)=WK(I) 00002920
000770 H=HOLD 00002930

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000771      NO=NOOLD      00002940
000772      JSTART=NO    00002950
000773      RETURN      00002960
*****00002970
C* THE CORRECTOR CONVERGED AND CONTROL IS PASSED TO STATEMENT 300  *00002980
C* IF THE ERROR TEST IS O.K., AND TO 310 OTHERWISE.             *00002990
C* IF THE STEP IS O.K. IT IS ACCEPTED. IF IDOUB HAS BEEN REDUCED *00003000
C* TO ONE, A TEST IS MADE TO SEE IF THE STEP CAN BE INCREASED   *00003010
C* AT THE CURRENT ORDER OR BY GOING TO ONE HIGHER OR ONE LOWER. *00003020
C* SUCH A CHANGE IS ONLY MADE IF THE STEP CAN BE INCREASED BY AT *00003030
C* LEAST 1.1. IF NO CHANGE IS POSSIBLE IDOUB IS SET TO 10 TU    *00003040
C* PREVENT FURTHER TESTING FOR 10 STEPS.                         *00003050
C* IF A CHANGE IS POSSIBLE, IT IS MADE AND IDOUB IS SET TO      *00003060
C* NO + 1 TO PREVENT FURTHER TESTING FOR THAT NUMBER OF STEPS.  *00003070
C* IF THE ERROR WAS TOO LARGE, THE OPTIMUM STEP SIZE FOR THIS OR *00003080
C* LOWER ORDER IS COMPUTED, AND THE STEP RETRIED. IF IT SHOULD  *00003090
C* FAIL TWICE MORE IT IS AN INDICATION THAT THE DERIVATIVES THAT *00003100
C* HAVE ACCUMULATED IN THE Y ARRAY HAVE ERRORS OF THE WRONG ORDER *00003110
C* SO THE FIRST DERIVATIVES ARE RECOMPUTED AND THE ORDER IS SET TO 1. *00003120
*****00003130
000774      285 D=0.      00003140
000775      DN 290 I=1,N 00003150
000777      290 D=D+(WK(N13+I)/WK(N12+I))**2 00003160
001010      THEVAL=0     00003170
001010      IF(D.GT.E)GOTO 310 00003180
001014      IF(K.LT.3)GOTO 300 00003190
*****00003200
C* COMPLETE THE CORRECTION OF THE HIGHER ORDER DERIVATIVES AFTER A *00003210
C* SUCCESSFUL STEP.      *00003220
*****00003230
001016      L=N          00003240
001017      DN 295 J=3,K 00003250
001020      L=L+N        00003260
001022      DN 295 I=1,N 00003270
001023      MN=L+I      00003280
001024      295 Y(MN)=Y(MN)+A(I)*WK(N13+I) 00003290
001037      300 IERR=+1  00003300
001040      HNEW=H      00003310
001041      IF(IDOUB.LE.1)GOTO 315 00003320
001044      IDOUB=IDOUB-1 00003330
001045      IF(IDOUB.GT.1)GOTO 390 00003340
001050      DN 305 I=1,N 00003350
001051      305 WK(N9+I)=WK(N13+I) 00003360

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001060 GOTO 390 00003370
C*****00003380
C* REDUCE THE FAILURE FLAG COUNT TO CHECK FOR MULTIPLE FAILURES. *00003390
C* RESTORE T TO ITS ORIGINAL VALUE AND TRY AGAIN UNLESS THERE HAVE *00003400
C* THREE FAILURES. IN THAT CASE THE DERIVATIVES ARE ASSUMED TO HAVE *00003410
C* ACCUMULATED ERRORS SO A RESTART FROM THE CURRENT VALUES OF Y IS *00003420
C* TRIED. *00003430
C*****00003440
001060 310 IERR=IERR-2 00003450
001062 IF(H.LE.HMIN*1.00001)GOTO 410 00003460
001066 T=TOLD 00003470
001067 IF(IERR.LE.-5)GOTO 400 00003480
C*****00003490
C* PR1, PR2, AND PR3 WILL CONTAIN THE AMOUNTS BY WHICH THE STEP SIZE *00003500
C* SHOULD BE DIVIDED AT ORDER ONE LOWER, AT THIS ORDER, AND AT ORDER *00003510
C* ONE HIGHER RESPECTIVELY. *00003520
C*****00003530
001071 315 PR2=(D/E)**ENQ2*1.2 00003540
001076 PR3=1.E+20 00003550
001100 IF(NO.GF.MAXDER.OR.IERR.LE.-1)GOTO 325 00003560
001112 D=0. 00003570
001113 DO 320 I=1,N 00003580
001114 320 D=D+((WK(N13+I)-WK(N9+I))/WK(N12+I))**2 00003590
001130 PR3=(D/EUP)**ENQ3*1.4 00003600
001135 325 PR1=1.E+20 00003610
001136 IF(NO.LE.1)GOTO 335 00003620
001141 D=0. 00003630
001142 NN=N*(K-1) 00003640
001145 DO 330 I=1,N 00003650
001146 NN=NN+1 00003660
001150 330 D=D+(Y(NN)/WK(N12+I))**2 00003670
001157 PR1=(D/EDWN)**ENQ1*1.3 00003680
001164 335 CONTINUE 00003690
001166 IF(IPR2.LE.PR3)GOTO 365 00003700
001167 IF(IPR3.LT.PR1)GOTO 370 00003710
001171 340 R=1./AMAX1(PR1,1.E-04) 00003720
001176 NE=Q-NO-1 00003730
001200 345 IDWUB=10 00003740
001201 IF(IERR.EQ.1.AND.R.LT.1.1)GOTO 390 00003750
001213 IF(NE=0.LE.NO)GOTO 355 00003760
C*****00003770
C* COMPUTE ONE ADDITIONAL SCALED DERIVATIVE IF ORDER IS INCREASED. *00003780
C*****00003790

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001215		NN=NEW0*N	00003800
001217		DO 350 I=1,N	00003810
001221		NN=NN+1	00003820
001223	350	Y(NN)=WK(N13+I)*A(K1/K	00003830
001233	355	K=NEWJ+1	00003840
001235		IF(ITER.EQ.1)GOTO 375	00003850
001237		RACUM=RACUM*R	00003860
001240		IRETI=3	00003870
001241		GOTO 415	00003880
001242	360	IF(NEW0.EQ.N0)GOTO 170	00003890
001244		N0=NEW0	00003900
001245		GOTO 55	00003910
001245	365	IF(PR2.GT.PR1)GOTO 340	00003920
001251		NEW0=N0	00003930
001251		R=1./AMAX1(PR2,1.E-04)	00003940
001256		GOTO 345	00003950
001257	370	R=1./AMAX1(PR3,1.E-04)	00003960
001264		NEW0=N0+1	00003970
001266		GOTO 345	00003980
001267	375	IRET=7	00003990
001270		R=AMINI(R,HMAX/ABS(H))	00004000
001275		H=H*R	00004010
001276		HNEW=H	00004020
001276		IF(N0.EQ.NEW0)GOTO 380	00004030
001301		N0=NEW0	00004040
001301		GOTO 55	00004050
001302	380	R1=1.	00004060
001303		L=0	00004070
001305		DO 385 J=2,K	00004080
001306		R1=R1*K	00004090
001310		L=L+N	00004100
001311		DO 385 I=1,N	00004110
001313		NN=L+1	00004120
001314	385	Y(NN)=Y(NN)*R1	00004130
001323		IDCUB=K	00004140
001324	390	DO 395 I=1,N	00004150
001326	395	WK(N12+I)=AMAX1(WK(N12+I),ABS(Y(I)))	00004160
001340		JSTART=NO	00004170
001341		RETURN	00004180
001342	400	IF(N0.EQ.1)GOTO 430	00004190
001346		CALL DIFFUN(T,Y,WK(N10P1))	00004200
001351		R=H/HOLD	00004210
001357		DO 405 I=1,N	00004220

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001360      Y(I)=WK(I)      00004230
001363      NN=N+1      00004240
001363      WK(NN)=HOLD*WK(N10+I) 00004250
001370      405 Y(NN)=WK(NN)*R 00004260
001376      NO=1      00004270
001376      IERR=+1      00004280
001400      GOTO 55      00004290
001401      410 IERR=-1      00004300
001402      HNEW=H      00004310
001403      JSTART=NO      00004320
001405      RETURN      00004330
C*****00004340
C* THIS SECTION SCALES ALL VARIABLES CONNECTED WITH H AND RETURNS *00004350
C* IN THE ENTERING SECTION. *00004360
C*****00004370
001406      415 RACUM=AMAX1(ABS(HMIN/HOLD),RACUM) 00004380
001414      RACUM=AMIN1(RACUM,ABS(HMAX/HOLD)) 00004390
001421      RI=1.      00004400
001422      L=0      00004410
001423      DO 420 J=2,K      00004420
001424      L=L+N      00004430
001425      RI=RI*RACUM      00004440
001427      DO 420 I=1,N      00004450
001431      NN=L+1      00004460
001432      420 Y(NN)=WK(NN)*RI      00004470
001442      H=HOLD*RACUM      00004480
001444      DO 425 I=1,N      00004490
001445      425 Y(I)=WK(I)      00004500
001452      IDOUB=K      00004510
001453      IF(I)ETI-2)20,170,360      00004520
001456      430 IERR=-4      00004530
001460      GOTO 275      00004540
001460      END      00004550

```

SUBROUTINE PEDERV(T,Y,PH,N)

- C
- C THIS IS A DUMMY SUBROUTINE TO SATISFY THE LOADER.
- C A DUMMY SUBROUTINE IS EMPLOYED WHEN NON-STIFF (MTH=0) METHODS ARE DESIRED.
- C

000007 RETURN
000010 END

SUBROUTINE MATENV(IPW,N,J)

C
C
C
C

THIS IS A DUMMY SUBROUTINE TO SATISFY THE LOADER.
A DUMMY SUBROUTINE IS EMPLOYED WHEN NON-STIFF (MTH=0) METHODS ARE DESIRED.

RETURN
END

000006
000007

SUBROUTINE DIFFUN(T,Y,DY)

C
C*****
C*
C* PURPOSE
C*
C* SUBROUTINE DIFFUN IS USED TO COMPUTE THE SCALED DERIVATIVES
C* USED BY DIFSUB. FLAGS MAY BE SET BY THE USER TO CHOOSE THE
C* DERIVATIVES FOR PENDULATION,TORSION,AND CORIOLIS EFFECTS IN
C* ANY COMBINATION DESIRED. THE INDIVIDUAL CONTRIBUTIONS ARE THEN
C* SUMMED TO OBTAIN A TOTAL COMPOSITE VALUE TO BE USED IN DIFSUB.
C*
C* PROPRIETARY
C*
C* NONE
C*
C* USE
C*
C* CALL DIFFUN(T,Y,DY)
C*
C* PARAMETERS
C*
C* T---TIME
C*
C* Y---ARRAY OF DEPENDENT VARIABLES AND THEIR SCALED DERIVATIVES
C*
C* DY---ARRAY OF THE DERIVATIVES OF Y
C*
C* REQUIRED ROUTINES
C*
C* PEND,TORS,CORL,DAMP
C*
C* AUTHOR/IMPLEMENTOR
C*
C* D.R.DORSEY,JR - B.SEAY
C*
C* DATE RELEASED
C*
C* SEPTEMBER 14,1973
C*
C* LATEST REVISION
C*

C-2

```
----- C*      SEPTEMBER 14,1973                               *
----- C*
----- C*****
----- C
----- C      I--- = 0 . OFF FLAG
----- C      I--- = 1 . ON FLAG
----- C
----- C*****
----- C
000006      COMMON/ TESTC/ IPEND,ITORS,ICORL
000006      DIMENSION Y(128),DY(16)
000006      DIMENSION DYP(16),DYT(16),DYC(16)
----- C
----- C*****
----- C
000006      DO 10 I=1,16
000007      DYP(I)=0.0
000010      DYT(I)=0.0
000011      DYC(I)=0.0
000012      10 CONTINUE
----- C
----- C
----- C--- PENDULATION MODULE
----- C
000014      IF(IPEND.EQ.0) GO TO 30
000015      CALL PEND(T,Y,DYP)
----- C
----- C--- TORSION MODULE
----- C
000016      30 IF(ITORS.EQ.0) GO TO 60
000021      CALL TORS(T,Y,DYT)
----- C
----- C--- CORIOLIS MODULE
----- C
000023      60 IF(ICORL.EQ.0) GO TO 90
000026      CALL CORL(T,Y,DYC)
000030      90 CONTINUE
----- C
----- C--- SUM UP CONTRIBUTIONS TO GET TOTAL DY - IF A MODULE WAS NOT
----- C--- USED,ITS CONTRIBUTION HAS BEEN PRESET TO ZERO.
----- C
000030      DO 100 I=1,16
000034      100 DY(I)=DYP(I)+DYT(I)+DYC(I)
```

.000042

C

RETURN

.000043

END

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SUBROUTINE PEND(T,Y,DY)

C
C*****
C*
C* PURPOSE
C*
C* SUBROUTINE PEND INTEGRATES THE DYNAMICALLY COUPLED LAGRANGIAN
C* EQUATIONS OF MOTION FOR THE PENDULATORY MOTION, USING THE
C* METHOD OF DETACHED COEFFICIENTS (SUBROUTINE BOSS) TO DECOUPLE
C* THE EQUATIONS BEFORE INTEGRATION. VISCOUS DAMPING MAY BE
C* INCLUDED, IF DESIRED.
C*
C* PROPRIETARY
C*
C* NONE
C*
C* USE
C*
C* CALL PEND(T,Y,DY)
C*
C* PARAMETERS
C*
C* T---TIME
C*
C* Y---ARRAY OF DEPENDENT VARIABLES AND SCALED DERIVATIVES
C*
C* DY--ARRAY OF DERIVATIVES OF Y WITH RESPECT TO T
C*
C* REQUIRED ROUTINES
C*
C* DAMP
C*
C* AUTHJR/IMPLEMENTOR
C*
C* D.R. DJRSEY, JR. - B. SEAY
C*
C* DATE RELEASED
C*
C* SEPTEMBER 14, 1973
C*
C* LATEST REVISION
C*

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```

C*   SEPTEMBER 14,1973   *
C*   *                   *
C*****
C
C
C*****
C
C
C*--*
C
C*****
C
C
C
C
000006 COMMON / CONST / MB,MS,MP,ME,M,RB,A1,L,P,RS,RP,RC,G,RHO,MU,N,
* HS,HP,HE
000006 COMMON / DAMPC / ASCS,APCP,AECE,D(18),IDAMP
000006 COMMON / SBSPE / SB,SS,SP,SE,SB1,SS1,SP1,SE1
000006 DIMENSION Y(128),M(4),DY(16),A(8,8),B(8),C(8,9),D(8)
000006 REAL MB,AS,MP,ME,M,L,MU
C
C*****
C
C
C COMPUTE THE THETAS AND PHIS
C
000006 THETAB=Y(1)
000007 PHIB=Y(2)
000010 THETAS=Y(3)
000012 PHIS=Y(4)
000013 THETAP=Y(5)
000015 PHIP=Y(6)
000016 THETAE=Y(7)
000020 PHIE=Y(8)
000021 TDOTB=Y(9)
000023 DDOTB=Y(10)
000024 TDOTS=Y(11)
000026 DDOTS=Y(12)
000027 TDOTP=Y(13)
000031 DDOTP=Y(14)
000032 TDOTE=Y(15)

```

```

-000034..... DATE=Y(16)
C
C
C*****
C
C      DRVROUTINE DIFPEN
C
C*****
C#
C# PURPOSE
C#
C#      DRVROUTINE DIFPEN DEFINES THE DERIVATIVES FOR PEND TO BE USED
C#      IN DIFSUB.
C#
C#
C# THIS PROGRAM COMPUTES THE DERIVATIVES OF Y(I) I=1,16 WITH RESPECT TO
C# T PLACES THE RESULTS IN DY(I) I=1,16 AND RETURNS TO DIFSUB
C#
C#
C#      INITIALIZE VALUES FOR DIFPEN
C#
000035..... XIB=39.369134*RB*RB
000037..... XIS=3.86955*PS*RS
000041..... XIP=1.2747*P*RP
000043..... XIE=7.329*RE*RL
C
000045..... Z1=XIB/(M(4)*RB*RB)
000050..... Z2=XIS/(M(3)*A1*A1)
000053..... Z3=XIP/(M(2)*L*L)
000056..... Z4=XIE/(M(1)*P*P)
C
000061..... SB=1./(Z1+1.)
000064..... SS=1./(Z2+1.)
000066..... SP=1./(Z3+1.)
000071..... SF=1./(Z4+1.)
C
000073..... Z1=Z1/(THE TAU*THETA B)
000075..... Z2=Z2/(THE TAU*THETA S)
000077..... Z3=Z3/(THE TAU*THETA P)
000100..... Z4=Z4/(THE TAU*THETA E)
C
000102..... SB1=1./(Z1+1.)
000105..... SS1=1./(Z2+1.)
000107..... SP1=1./(Z3+1.)
000112..... SF1=1./(Z4+1.)

```

C
C
C
C
FORM THE A MATRIX

000114 DD 10 I=1,N

000116 10 A(1,1)=1.0

000124 A(1,2)=A(1,4)=A(2,1)=A(3,2)=A(3,4)=A(4,3)=A(5,6)=A(5,8)=A(6,5)=

1A(7,6)=A(7,8)=A(8,7)=0.0

C
C
C
C
SET UP THE INITIAL VALUES

000162 C1=SB/(M(4)*RB)

000165 D1=M(3)*A1

000167 E1=M(2)*L

000171 F1=M(1)*P

000173 G1=RB*THETA

000175 H1=A1*THETA

000177 H2=RB*PHI

000201 A(1,3)=D1*C1

000203 A(1,5)=F1*C1

000204 A(1,6)=F1*THETA*(PHI-PHI)*C1

000211 A(1,7)=F1*C1

000212 A(1,8)=F1*THETA*(PHI-PHI)*C1

C
C
C
C
RESET C1

000217 C1=SB1/(M(4)*G1)

000222 A(2,3)=D1*PHI*C1

000224 A(2,4)=D1*THETA*C1

000226 A(2,5)=E1*PHI*C1

000230 A(2,6)=E1*THETA*C1

000232 A(2,7)=F1*PHI*C1

000234 A(2,8)=F1*THETA*C1

C
C
C
C
RESET C1

000236 C1=SS/A1

000240 A(3,1)=RB*C1

000242 C1=C1/M(3)

000244 A(3,5)=E1*C1

000245 A(3,6)=E1*THETA*(PHI-PHI)*C1

000252 A(3,7)=F1*C1

000253 A(3,8)=F1*THETA*(PHI-PHI)*C1

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C
000334 C1=SE/P
000336 A(7,1)=R1*C1
000337 A(7,2)=G1*(PHIE-PHIB)*C1
000342 A(7,4)=H1*(PHIE-PHIS)*C1
000346 A(7,3)=A1*C1
000347 A(7,5)=L*C1

C
RESFT C1

C
000351 C1=SE1/(P*THETA)
000354 A(8,1)=H2*C1
000355 A(8,2)=G1*C1
000357 A(8,3)=A1*PHIS*C1
000361 A(8,4)=H1*C1
000362 A(8,5)=L*PHIP*C1
000365 A(8,6)=L*THETA*P*C1

C
FORM THE B-MATRIX

C
SET UP INITIAL CONDITIONS

C
000367 P1B2TB=DDTB*COA*THETAB
000371 P1S2TS=DDTS*DOTS*THETAS
000373 P1P2TP=DDTP*DOTP*THETAP
000375 P1E2TE=DDTE*DOTE*THETA

C
000377 P1H2DB=DDTB*TOUB
000401 P1S2DS=DDTS*TOUS
000403 P1P2DP=DDTP*TOUP
000405 P1E2DE=DDTE*TOUE

C
000407 P1MPP=PHIB-PHIP
000411 P1MPE=PHIB-PHIE
000413 P1SMP=PHIS-PHIP
000414 P1SMP=PHIS-PHIE

C
000416 TBMTB=THETAB-THETA
000420 TBMTS=THETAB-THETA
000422 TSMTP=THETAS-THETA
000423 TSMTE=THETAS-THETA

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C
C
C
000260 C1=SS1/H1
000262 A(4.1)=H2*C1
000263 A(4.2)=G1*C1
C
C
C
000265 C1=C1/M(3)
000267 A(4.5)=E1*PHIP*C1
000271 A(4.6)=E1*THETAP*C1
000273 A(4.7)=F1*PHIE*C1
000275 A(4.8)=F1*THETAEC*C1
C
C
C
000277 C1=SP/L
000301 A(5.1)=RB*C1
000302 A(5.2)=G1*(PHIP-PHIB)*C1
000305 A(5.3)=R1*C1
000307 A(5.4)=H1*(PHIP-PHIS)*C1
C
C
C
000312 C1=C1/M(2)
000313 A(5.7)=F1*C1
C
C
C
000315 C1=SP1/(L*THETAP)
000320 A(6.1)=H2*C1
000321 A(6.2)=G1*C1
000323 A(6.3)=A1*PHIS*C1
000325 A(6.4)=H1*C1
C
C
C
000326 C1=C1/M(2)
000330 A(6.7)=F1*PHIE*C1
000332 A(6.8)=F1*THETAEC*C1
C
C

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```

000425- R1=1./M(4)*RB)
000430 R2=M(2)*L
000432 R3=M(1)*P
000434 R4=M(3)*A1
----- C
----- C R1, Q1, AND W1 WILL VARY WHEN NECESSARY
----- C
----- C COMPUTE THE B-MATRIX
----- C
000436 B(1)=SB*(PUB2TB-R1*(M(3)*(-A1*PDS2TS+G*THETAB)
1 +R2*(TDOTP*(2.*DOTP*PUMPP+TDOTP*THMTP)-PDP2TP)
2 +R3*(TDOTE*(2.*DOTE*PBMPE+TDOTE*TBMTF)-PDF2TE)))
----- C
000466 Q1=-2./THETAB
----- C
000470 B(2)=Q1*SH1*(R1*(R4*PDSTDS+R2*PDPTDP+R3*PDEIDE)+PBDTDB)
----- C
000501 R1=1./A1
000503 W1=1./M(3)
----- C
000505 B(3)=SS*(POS2TS-R1*((-RB*PUB2TB+G*THETAS)
1 +W1*(R2*(TDOTP*(2.*DOTP*PSMPP+TDOTP*TSMTF)-PDP2TP)
2 +R3*(TDOTE*(2.*DOTE*PSMPE+TDOTE*TSMTF)-PDE2TE)))
----- C
000535 Q1=(Q1*THETAB)/THETAS
000540 R1=R1*W1
----- C
000541 B(4)=Q1*SS1*(R1*(M(3)*RB*PUBTDB+R2*PDPTDP+R3*PDEIDE)+PDSTDS)
----- C
000553 R1=1./L
000555 W1=1./M(2)
----- C
000557 B(5)=SP*(PDP2TP-R1*(RB*(TDOTB*(2.*DOTB*(-PRMPP)+TDOTB*(-TBMTF))-
1 PUB2TB)
2 +A1*(TDOTS*(2.*DOTS*(-PSMPP)+TDOTS*(-TSMTF)-PDS2TS)
3 +G*THETAP-R3*W1*PDE2TE)))
----- C
000612 Q1=(Q1*THETAS)/THETAP
000614 R1=R1*W1
----- C
000616 B(6)=Q1*SP1*(R1*(M(2)*RB*PUBTDB+A1*PDSTDS)+R3*PDEIDE)+PDPTDP)
----- C
000630 R1=1./P

```

C
000631 BL7)=SE*(PDE2TE-R1*(RB*(TDO7B*(2.*DOTB*(-PEMPE)+TDO7B*(-TBME))-
1 - PDE2TB)
2 +A1*(TDO7S*(2.*DOTS*(-PSMPE)+TDO7S*(-TSMTE))-PDS2TS)
3 -L*PDP2TP+G*THETA)

C
000663 Q1=(Q1+THETA)/THETA

C
000665 B(8)=Q1*SE1*(R1*(RB*PDBTDB+A1*PDS2DS+L*PDPTDP)+PDETDE)

C
C ADD IN DAMPING CONTRIBUTION

C
000676 IF(IIDAMP.EQ.0) GO TO 30

000677 CALL DAMP(Y)

000701 DO 20 K=1,8

000705 20 B(K)=B(K)+D(K)

C
000711 30 CONTINUE

C
C FORM THE C-MATRIX

C
000711 DO 40 I=1,8

000713 C(I,9)=B(I)

000715 DO 40 J=1,8

000716 40 C(I,J)=A(I,J)

000730 CALL BOSS(C,N,0)

C
C THE DERIVATIVES ARE FORMED AS FOLLOWS

C
000732 DO 50 I=1,8

000736 DY(I)=Y(I+8)

000741 DY(I+8)=D(I)

000742 50 CONTINUE

C
C

000744 RETURN

000745 END

SUBROUTINE TORS(T,Y,DY)

C
C*****
C*
C* PURPOSE
C*
C* SUBROUTINE TORS INTEGRATES THE TORSIONAL EQUATIONS OF MOTION.
C* DAMPING IS ALREADY INCLUDED. THESE EQUATIONS HAVE ONLY STATIC
C* COUPLING AND CAN BE INTEGRATED DIRECTLY BY DIFSUB.
C*
C* PROPRIETARY
C*
C* NONE
C*
C* USE
C*
C* CALL TORS(T,Y,DY)
C*
C* PARAMETERS
C*
C* I---TIME
C*
C* Y---ARRAY OF DEPENDENT VARIABLES AND SCALED DERIVATIVES
C*
C* DY---ARRAY OF DERIVATIVES OF Y WITH RESPECT TO T
C*
C* REQUIRED ROUTINES
C*
C* NONE
C*
C* AUTHOR/IMPLEMENTOR
C*
C* D. R. DURSEY, JR - B. SEAY
C*
C* DATE RELEASED
C*
C* SEPTEMBER 14, 1973
C*
C* LATEST REVISION
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C* SEPTEMBER 14, 1973
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C*****
C
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C*--*
C
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C*****
C
C
C
000006 COMMON / CONST / MB,MS,MP,ME,M,RB,A1,L,P,RS,RP,RE,G,RHO,MU,N,
* HS,HP,HE
000006 REAL MB,MS,MP,ME,M,L,MU
000006 DIMENSION Y(34),M(4),DY(16)
C
C*****
C
000006 U=RB+A1
000007 TAUS=MS*G*KS*RS/O
000013 Q=O+L+HS
000016 TAUP=MP*G*RP*RP/Q
000022 Q=U+P+HP
000024 TAU=ME*G*RE*RE/Q
C
C
000030 DN=2.5*SQRT(RHO*MU)
000035 ETAB=DN*RB**4*SQRT(ABS(Y(10)))
000047 FTAS=DN*RS**4*SQRT(ABS(Y(12)))
000057 ETAP=DN*RP**4*SQRT(ABS(Y(14)))
000067 ETAE=DN*RE**4*SQRT(ABS(Y(16)))
C
000077 AIB=2.*MB*RB*RP/3.
000103 AIS=0.5*MS*RS*KS
000106 AIP=0.5*MP*RP*RP
000110 AIE=0.5*ME*RE*RE
C
C DBVRoutine DIFTOR
C
C*****

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```

C*
C* PURPOSE
C*
C*   DRVROUTINE DIFTOR DEFINES THE DERIVATIVES FOR TORS TO BE USED
C*   IN DIFSUB.
C*
C*
C*****
C
C
C   THE DERIVATIVES ARE COMPUTED AS FOLLOWS
C
C
.000113   DY(1) =0.0
.000114   DY(2) =Y(10)
.000115   DY(3) =0.0
.000116   DY(4) =Y(12)
.000120   DY(5) =0.0
.000121   DY(6) =Y(14)
.000122   DY(7) =0.0
.000123   DY(8) =Y(16)
.000125   DY(9) =0.0
.000126   DY(10)=1./A14*(-ETAB*Y(10)-TAUS*(Y(2)-Y(4)))
.000134   DY(11)=0.0
.000135   DY(12)=1./A15*(-ETAS*Y(12)+TAUS*(Y(2)-Y(4))-TAUP*(Y(4)-Y(6)))
.000150   DY(13)=0.0
.000151   DY(14)=1./A1P*(-ETAP*Y(14)+TAUP*(Y(4)-Y(6))-TAUE*(Y(6)-Y(8)))
.000164   DY(15)=0.0
.000165   DY(16)=1./A1C*(-ETAEC*Y(16)+TAUE*(Y(6)-Y(8)))
C
C
.000175   RETURN
.000175   END

```

SUBROUTINE CORL(T,Y,DY)

C*****
C*
C* PURPOSE
C* SUBROUTINE CORL INTEGRATES THE 2-D EQUATIONS OF MOTION FOR THE
C* CORIOLIS EFFECT. THE EQUATIONS HAVE ONLY STATIC COUPLING AND
C* CAN BE INTEGRATED DIRECTLY BY DIFSUB. NO DAMPING IS INCLUDED.
C*
C* PROPRIETARY
C*
L* NONE
C*
C* USE
C*
C* CALL CORL(T,Y,DY)
C*
C* PARAMETERS
C*
L* T---TIME
C*
C* Y---ARRAY OF DEPENDENT VARIABLES AND SCALED DERIVATIVES
C*
C* DY---ARRAY OF DERIVATIVES OF Y WITH RESPECT TO T
C*
L* REQUIRED ROUTINES
C*
C* NONE
C*
C* AUTHOR/IMPLEMENTOR
C*
C* D.R.DORSEY,JR - B.SEAY
C*
C* DATE RELEASED
C*
C* SEPTEMBER 14,1973
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000006 COMMON/ CORIO / OMEGA,LAMDA
000006 COMMON / CONST / MB,MS,MP,ME,M,RB,AL,L,P,RS,RP,RE,G,RHO,MU,N,
      * HS,HP,HE
000006 DIMENSION Y(128),M(4),DY(16)
000006 REAL MB,MS,MP,ME,M,L,MU,LAMDA
000006 REAL MUB,MUS,MUP,MUE

```

```

C
C *****
C
C SET UP XI≠S AND MU≠S
000006 XIB=Y(9)*Y(2)+Y(10)*Y(1)
000011 XIS=Y(11)*Y(4)+Y(12)*Y(3)
000016 XIP=Y(13)*Y(6)+Y(14)*Y(5)
000022 XIE=Y(15)*Y(8)+Y(16)*Y(7)
000027 MUB=Y(9)-Y(10)*Y(1)*Y(2)
000033 MUS=Y(11)-Y(12)*Y(3)*Y(4)
000040 MUP=Y(13)-Y(14)*Y(5)*Y(6)
000044 MUE=Y(15)-Y(16)*Y(7)*Y(8)

```

```

C
C SET UP INITIAL VALUES
C
000051 T1=2.*MB*OMEGA*RB*RB
000055 SLAM=2.*SIN(LAMDA)
000060 CLAM=COS(LAMDA)

```

```

C
C DRVRoutine DIFCOR

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C *****
C# PURPOSE
C# ORVRoutine DIFCOR DEFINES THE DERIVATIVES FOR CORL TO BE USED
C# IN DIFSUB.
C
C THIS PROGRAM COMPUTES THE DERIVATIVES OF Y(I) I=1,16 WITH RESPECT TO
C T. PLACES THE RESULTS IN DY(I) I=1,16 AND RETURNS TO DIFSUB
C
C SET UP THE FIRST DERIVATIVES
C
000063 DO 10 I=1,9
000066 DY(I)=Y(I+d)
000070 10 CONTINUE
C
000072 DY(9) =T1*(SLAM*(XIB-MUB*Y(2))+CLAM*XIB*Y(1))
000102 DY(10)=T1*(-SLAM*(XIB*Y(2)+MUB)*Y(1))
C
C RESET THE INITIAL VALUES
C
000107 T1=2.*(MB+MS)*OMEGA
000113 T2=T1*A1
000114 XT1=RB*XIB+A1*XIS
000120 XMU1=RB*MUB+A1*MUS
C
C COMPUTE MORE DERIVATIVES
C
000123 DY(11)=T2*(SLAM*(XI1-XMU1*Y(4))+CLAM*XI1*Y(3))
000133 DY(12)=T2*(-SLAM*(XI1*Y(4)+XMU1)*Y(3))
C
C RESET VALUES
C
000141 T1=T1+2.*MP*OMEGA
000144 T2=T1*L
000146 XI1=XI1+L*XIP
000151 XMU1=XMU1+L*MUP
C
C COMPUTE FURTHER DY#S
C
000154 DY(13)=T2*(SLAM*(XI1-XMU1*Y(6))+CLAM*XI1*Y(5))

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000164..... DY(14)=T2*(-SLAM*(X11*Y(6)+XMUI)*Y(5))
..... C
..... C   RESET LAST INITIAL VALUES
..... C
000172..... T1=T1+2.*ME*OMEGA
000175..... T2=T1*P
000177..... XII=X11+P*XIE
000202..... XMUI=XMUI+P*MUE
..... C
..... C   COMPUTE LAST SET OF DERIVATIVES
..... C
000205..... DY(15)=T2*(SLAM*(X11-XMUI*Y(6))+CLAM*X11*Y(7))
000215..... DY(16)=T2*(-SLAM*(X11*Y(8)+YMUI)*Y(7))
..... C
..... C
000223..... RETURN
000224..... END
```

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SUBROUTINE DAMP(Y)

C
C*****
C#
C# PURPOSE
C#
C# SUBROUTINE DAMP DEFINES THE VISCOUS/AERODYNAMIC DAMPING EFFECTS
C# THAT MAY BE INCLUDED IN SUBROUTINE PEND. THE DAMPING VALUES
C# ARE RETURNED TO PEND THROUGH THE ARRAY D(1)-D(8).
C#
C# PROPRIETARY
C#
L# NONE
C#
C# USE
C#
C# CALL DAMP(Y)
C#
C# PARAMETERS
C#
C# Y---ARRAY OF DEPENDENT VARIABLES AND SCALED DERIVATIVES
C#
C# REQUIRED ROUTINES
C#
C# NONE
C#
C# AUTHOR/IMPLEMENTOR
C#
C# D.R.DORSEY, JR - B. SEAY
C#
C# DATE RELEASED
C#
C# SEPTEMBER 14, 1973
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000003 COMMON / CONST / MB,MS,MP,ME,M,RB,A1,L,P,RS,RP,RE,G,RHO,MU,N,
* HS,HP,HE
000003 COMMON / DAMPC / ASCS,APCP,AECE,D(8),IDAMP
000003 COMMON / SBSPE / SB,SS,SP,SE,SBI,SSI,SP1,SE1
000003 RFAL MB,MS,MP,ME,M,L,MU
000003 DIMENSION M(4),Y(12B)

```

```

C*****
C
C INITIALIZE VALUES FOR B-MATRIX
C

```

```

000003 RHO=2.5*SQRT(RHO*MU)
000010 ATDDB=ABS(Y(9))
000013 SATDB=SQRT(ATDDB)
000015 D(1)=RHO*RB*RB*SATDB*Y(9)/M(4)*SB

```

```

C
000024 ADDB=ABS(Y(10))
000026 SAAB=SQRT(ADDB)
000030 D(2)=RHO*RB*RB*SAAB*Y(10)/(M(4)*Y(1)*Y(11)*SBI

```

```

C
000040 RAC=0.5*RHO*ASCS
000042 Z1=RB+A1
000044 X11=RB*Y(9)+A1*Y(11)
000050 AX11=ABS(X11)
000051 X12=AX11*X11
000053 D(3)=RAC*Z1*X12/(M(3)*A1*A1)*SS

```

```

C
000060 Y11=RB*Y(1)*Y(10)+A1*Y(3)*Y(12)
000065 AY11=ABS(Y11)
000067 Y12=AY11*Y11
000071 D(4)=RAC*Z1*Y12/(M(3)*A1*A1*Y(3)*Y(3))*SSI

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C

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```

000100 RAC=0.5*RHO*APCP
000102 Z1=Z1+L
000104 X11=X11+L*Y(13)
000107 AX11=ABS(X11)
000110 X12=AX11*X12
000112 D(5)=RAC*Z1*X12/(M(2)*L*L)*SP
C
000117 Y11=Y11+L*Y(14)*Y(5)
000123 AY11=ABS(Y11)
000124 Y12=AY11*Y11
000125 D(6)=RAC*Z1*Y12/(M(2)*L*L*Y(5)*Y(5))*SP1
C
000134 RAC=0.5*RHO*AECE
000136 Z1=Z1+P
000140 X11=X11+P*Y(15)
000143 AX11=ABS(X11)
000144 X12=AX11*X11
000146 D(7)=RAC*Z1*X12/(M(1)*P*P)*SE
C
000153 Y11=Y11+P*Y(16)*Y(7)
000157 AY11=ABS(Y11)
000160 Y12=AY11*Y11
000161 D(8)=RAC*Z1*Y12/(M(1)*L*L*Y(7)*Y(7))*SE1
C
000170 RETURN
000171 END

```

SUBROUTINE BOSS(A,N,D)

BOSS0001

C
C* *****
C* PURPOSE
C* SUBROUTINE BOSS EMPLOYS THE METHOD OF DETACHED COEFFICIENTS TO
C* SOLVE A SYSTEM OF REAL LINEAR EQUATIONS REPRESENTED BY THE MATRIX*
C* EQUATION $AX=B$, WHERE A IS A SQUARE MATRIX OF COEFFICIENTS, X IS*
C* THE SOLUTION VECTOR, AND B IS THE RHS CONSTANT VECTOR.
C* PROPRIETARY
C* NONE
C* USE
C* CALL BOSS(A,N,D)
C* PARAMETERS
C* A AN AUGMENTED INPUT MATRIX CONTAINING THE LHS MATRIX OF
C* COEFFICIENTS IN THE FIRST N COLUMNS AND THE RHS CONSTANT
C* VECTOR IN THE (N+1)-ST COLUMN. THIS MATRIX MUST BE
C* DIMENSIONED AT LEAST (N,N+1) IN THE CALLING PROGRAM.
C* N THE NUMBER OF ROWS IN THE A MATRIX.
C* D THE SOLUTION VECTOR. THIS VECTOR MUST BE DIMENSIONED AT
C* LEAST N IN THE CALLING PROGRAM.
C* REQUIRED ROUTINES
C* NONE
C* AUTHOR/IMPLEMENTOR
C* V.C.TAYLOR
C* DATE RELEASED
C* JUNE 14, 1973

C*
C* LATEST REVISION

C*
C* JUNE 14, 1973

C*
C*****

C
C

.000006	DIMENSION A(N,1),O(N)	BOSS0002
.000006	B12=A(1,2)/A(1,1)	BOSS0003
.000012	B13=A(1,3)/A(1,1)	BOSS0004
.000015	B14=A(1,4)/A(1,1)	BOSS0005
.000021	B15=A(1,5)/A(1,1)	BOSS0006
.000024	B16=A(1,6)/A(1,1)	BOSS0007
.000030	B17=A(1,7)/A(1,1)	BOSS0008
.000033	B18=A(1,8)/A(1,1)	BOSS0009
.000037	B19=A(1,9)/A(1,1)	BOSS0010
.000042	B22=(A(2,2)-A(2,1)*B12)	BOSS0011
.000051	B23=(A(2,3)-A(2,1)*B13)	BOSS0012
.000061	B24=(A(2,4)-A(2,1)*B14)	BOSS0013
.000070	B25=(A(2,5)-A(2,1)*B15)	BOSS0014
.000100	B26=(A(2,6)-A(2,1)*B16)	BOSS0015
.000107	B27=(A(2,7)-A(2,1)*B17)	BOSS0016
.000117	B28=(A(2,8)-A(2,1)*B18)	BOSS0017
.000126	B29=(A(2,9)-A(2,1)*B19)	BOSS0018
.000136	B32=(A(3,2)-A(3,1)*B12)	BOSS0019
.000145	B33=(A(3,3)-A(3,1)*B13)	BOSS0020
.000155	B34=(A(3,4)-A(3,1)*B14)	BOSS0021
.000164	B35=(A(3,5)-A(3,1)*B15)	BOSS0022
.000174	B36=(A(3,6)-A(3,1)*B16)	BOSS0023
.000203	B37=(A(3,7)-A(3,1)*B17)	BOSS0024
.000213	B38=(A(3,8)-A(3,1)*B18)	BOSS0025
.000222	B39=(A(3,9)-A(3,1)*B19)	BOSS0026
.000232	B42=(A(4,2)-A(4,1)*B12)	BOSS0027
.000241	B43=(A(4,3)-A(4,1)*B13)	BOSS0028
.000251	B44=(A(4,4)-A(4,1)*B14)	BOSS0029
.000260	B45=(A(4,5)-A(4,1)*B15)	BOSS0030
.000270	B46=(A(4,6)-A(4,1)*B16)	BOSS0031
.000277	B47=(A(4,7)-A(4,1)*B17)	BOSS0032
.000307	B48=(A(4,8)-A(4,1)*B18)	BOSS0033
.000316	B49=(A(4,9)-A(4,1)*B19)	BOSS0034
.000326	B52=(A(5,2)-A(5,1)*B12)	BOSS0035
.000335	B53=(A(5,3)-A(5,1)*B13)	BOSS0036

000345	B54=(A(5.4)-A(5.1)*B14)	BOSS0037
000354	B55=(A(5.5)-A(5.1)*B15)	BOSS0038
000364	B56=(A(5.6)-A(5.1)*B16)	BOSS0039
000373	B57=(A(5.7)-A(5.1)*B17)	BOSS0040
000403	B58=(A(5.8)-A(5.1)*B18)	BOSS0041
000412	B59=(A(5.9)-A(5.1)*B19)	BOSS0042
000422	B62=(A(6.2)-A(6.1)*B12)	BOSS0043
000431	B63=(A(6.3)-A(6.1)*B13)	BOSS0044
000441	B64=(A(6.4)-A(6.1)*B14)	BOSS0045
000450	B65=(A(6.5)-A(6.1)*B15)	BOSS0046
000460	B66=(A(6.6)-A(6.1)*B16)	BOSS0047
000467	B67=(A(6.7)-A(6.1)*B17)	BOSS0048
000477	B68=(A(6.8)-A(6.1)*B18)	BOSS0049
000506	B69=(A(6.9)-A(6.1)*B19)	BOSS0050
000516	B72=(A(7.2)-A(7.1)*B12)	BOSS0051
000525	B73=(A(7.3)-A(7.1)*B13)	BOSS0052
000535	B74=(A(7.4)-A(7.1)*B14)	BOSS0053
000544	B75=(A(7.5)-A(7.1)*B15)	BOSS0054
000554	B76=(A(7.6)-A(7.1)*B16)	BOSS0055
000563	B77=(A(7.7)-A(7.1)*B17)	BOSS0056
000573	B78=(A(7.8)-A(7.1)*B18)	BOSS0057
000602	B79=(A(7.9)-A(7.1)*B19)	BOSS0058
000612	B82=(A(8.2)-A(8.1)*B12)	BOSS0059
000621	B83=(A(8.3)-A(8.1)*B13)	BOSS0060
000631	B84=(A(8.4)-A(8.1)*B14)	BOSS0061
000640	B85=(A(8.5)-A(8.1)*B15)	BOSS0062
000650	B86=(A(8.6)-A(8.1)*B16)	BOSS0063
000657	B87=(A(8.7)-A(8.1)*B17)	BOSS0064
000667	B88=(A(8.8)-A(8.1)*B18)	BOSS0065
000676	B89=(A(8.9)-A(8.1)*B19)	BOSS0066
000706	C23=B23/B22	BOSS0067
000710	C24=B24/B22	BOSS0068
000711	C25=B25/B22	BOSS0069
000713	C26=B26/B22	BOSS0070
000714	C27=B27/B22	BOSS0071
000716	C28=B28/B22	BOSS0072
000720	C29=B29/B22	BOSS0073
000721	C13=(B13-B12*C23)	BOSS0074
000724	C14=(B14-B12*C24)	BOSS0075
000727	C15=(B15-B12*C25)	BOSS0076
000732	C16=(B16-B12*C26)	BOSS0077
000735	C17=(B17-B12*C27)	BOSS0078
000740	C18=(B18-B12*C28)	BOSS0079

000743	C19=(B19-B12*C29)	BOSS0080
000746	C33=(B33-B32*C23)	BOSS0081
000751	C34=(B34-B32*C24)	BOSS0082
000754	C35=(B35-B32*C25)	BOSS0083
000757	C36=(B36-B32*C26)	BOSS0084
000762	C37=(B37-B32*C27)	BOSS0085
000765	C38=(B38-B32*C28)	BOSS0086
000770	C39=(B39-B32*C29)	BOSS0087
000773	C43=(B43-B42*C23)	BOSS0088
000776	C44=(B44-B42*C24)	BOSS0089
001001	C45=(B45-B42*C25)	BOSS0090
001004	C46=(B46-B42*C26)	BOSS0091
001007	C47=(B47-B42*C27)	BOSS0092
001012	C48=(B48-B42*C28)	BOSS0093
001015	C49=(B49-B42*C29)	BOSS0094
001020	C53=(B53-B52*C23)	BOSS0095
001023	C54=(B54-B52*C24)	BOSS0096
001026	C55=(B55-B52*C25)	BOSS0097
001031	C56=(B56-B52*C26)	BOSS0098
001034	C57=(B57-B52*C27)	BOSS0099
001037	C58=(B58-B52*C28)	BOSS0100
001042	C59=(B59-B52*C29)	BOSS0101
001045	C63=(B63-B62*C23)	BOSS0102
001050	C64=(B64-B62*C24)	BOSS0103
001053	C65=(B65-B62*C25)	BOSS0104
001056	C66=(B66-B62*C26)	BOSS0105
001061	C67=(B67-B62*C27)	BOSS0106
001064	C68=(B68-B62*C28)	BOSS0107
001067	C69=(B69-B62*C29)	BOSS0108
001072	C73=(B73-B72*C23)	BOSS0109
001075	C74=(B74-B72*C24)	BOSS0110
001100	C75=(B75-B72*C25)	BOSS0111
001103	C76=(B76-B72*C26)	BOSS0112
001106	C77=(B77-B72*C27)	BOSS0113
001111	C78=(B78-B72*C28)	BOSS0114
001114	C79=(B79-B72*C29)	BOSS0115
001117	C83=(B83-B82*C23)	BOSS0116
001122	C84=(B84-B82*C24)	BOSS0117
001125	C85=(B85-B82*C25)	BOSS0118
001130	C86=(B86-B82*C26)	BOSS0119
001133	C87=(B87-B82*C27)	BOSS0120
001136	C88=(B88-B82*C28)	BOSS0121
001141	C89=(B89-B82*C29)	BOSS0122

001144 D34=C34/C33
 001146 D35=C35/C33
 001150 D36=C36/C33
 001151 D37=C37/C33
 001153 D38=C38/C33
 001154 D39=C39/C33
 001156 D14=(C14-C13*034)
 001161 D15=(C15-C13*035)
 001164 D16=(C16-C13*036)
 001167 D17=(C17-C13*037)
 001172 D18=(C18-C13*038)
 001175 D19=(C19-C13*039)
 001200 D24=(C24-C23*034)
 001203 D25=(C25-C23*035)
 001206 D26=(C26-C23*036)
 001211 D27=(C27-C23*037)
 001214 D28=(C28-C23*038)
 001217 D29=(C29-C23*039)
 001222 D44=(C44-C43*034)
 001225 D45=(C45-C43*035)
 001230 D46=(C46-C43*036)
 001233 D47=(C47-C43*037)
 001236 D48=(C48-C43*038)
 001241 D49=(C49-C43*039)
 001244 D54=(C54-C53*034)
 001247 D55=(C55-C53*035)
 001252 D56=(C56-C53*036)
 001255 D57=(C57-C53*037)
 001260 D58=(C58-C53*038)
 001263 D59=(C59-C53*039)
 001266 D64=(C64-C63*034)
 001271 D65=(C65-C63*035)
 001274 D66=(C66-C63*036)
 001277 D67=(C67-C63*037)
 001302 D68=(C68-C63*038)
 001305 D69=(C69-C63*039)
 001310 D74=(C74-C73*034)
 001313 D75=(C75-C73*035)
 001316 D76=(C76-C73*036)
 001321 D77=(C77-C73*037)
 001324 D78=(C78-C73*038)
 001327 D79=(C79-C73*039)
 001332 D84=(C84-C83*034)

80SS0123
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 80SS0125
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 80SS0165

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001335	D85=(C85-C83*D35)	BOSS0166
001340	D86=(C86-C83*D36)	BOSS0167
001343	D87=(C87-C83*037)	BOSS0168
001346	D88=(C88-C83*038)	BOSS0169
001351	D89=(C89-C83*039)	BOSS0170
001354	E45=D45/D44	BOSS0171
001356	E46=D46/D44	BOSS0172
001360	E47=D47/D44	BOSS0173
001361	E48=D48/D44	BOSS0174
001363	E49=D49/D44	BOSS0175
001364	E15=(015-D14*E45)	BOSS0176
001367	E16=(016-D14*E46)	BOSS0177
001372	E17=(017-D14*E47)	BOSS0178
001375	E18=(018-D14*E48)	BOSS0179
001400	E19=(019-D14*E49)	BOSS0180
001403	E25=(025-D24*E45)	BOSS0181
001406	E26=(026-D24*E46)	BOSS0182
001411	E27=(027-D24*E47)	BOSS0183
001414	E28=(028-D24*E48)	BOSS0184
001417	E29=(029-D24*E49)	BOSS0185
001422	E35=(035-D34*E45)	BOSS0186
001425	E36=(036-D34*E46)	BOSS0187
001430	E37=(037-D34*E47)	BOSS0188
001433	E38=(038-D34*E48)	BOSS0189
001436	E39=(039-D34*E49)	BOSS0190
001441	E55=(055-D54*E45)	BOSS0191
001444	E56=(056-D54*E46)	BOSS0192
001447	E57=(057-D54*E47)	BOSS0193
001452	E58=(058-D54*E48)	BOSS0194
001455	E59=(059-D54*E49)	BOSS0195
001460	E65=(065-D64*E45)	BOSS0196
001463	E66=(066-D64*E46)	BOSS0197
001466	E67=(067-D64*E47)	BOSS0198
001471	E68=(068-D64*E48)	BOSS0199
001474	E69=(069-D64*E49)	BOSS0200
001477	E75=(075-D74*E45)	BOSS0201
001502	E76=(076-D74*E46)	BOSS0202
001505	E77=(077-D74*E47)	BOSS0203
001510	E78=(078-D74*E48)	BOSS0204
001513	E79=(079-D74*E49)	BOSS0205
001516	E85=(085-D84*E45)	BOSS0206
001521	E86=(086-D84*E46)	BOSS0207
001524	E87=(087-D84*E47)	BOSS0208

001527	E88=(D88-D84*E48)	BOSS0209
001537	E89=(D89-D84*E49)	BOSS0210
001536	F56=E56/E55	BOSS0211
001537	F57=E57/E55	BOSS0212
001541	F58=E58/E55	BOSS0213
001542	F59=E59/E55	BOSS0214
001544	F16=(E16-E15*F56)	BOSS0215
001547	F17=(E17-E15*F57)	BOSS0216
001552	F18=(E18-E15*F58)	BOSS0217
001552	F19=(E19-E15*F59)	BOSS0218
001560	F26=(E26-E25*F56)	BOSS0219
001563	F27=(E27-E25*F57)	BOSS0220
001566	F28=(E28-E25*F58)	BOSS0221
001571	F29=(E29-E25*F59)	BOSS0222
001574	F36=(E36-E35*F56)	BOSS0223
001577	F37=(E37-E35*F57)	BOSS0224
001602	F38=(E38-E35*F58)	BOSS0225
001605	F39=(E39-E35*F59)	BOSS0226
001610	F46=(E46-E45*F56)	BOSS0227
001613	F47=(E47-E45*F57)	BOSS0228
001616	F48=(E48-E45*F58)	BOSS0229
001621	F49=(E49-E45*F59)	BOSS0230
001624	F66=(E66-E65*F56)	BOSS0231
001627	F67=(E67-E65*F57)	BOSS0232
001632	F68=(E68-E65*F58)	BOSS0233
001635	F69=(E69-E65*F59)	BOSS0234
001640	F76=(E76-E75*F56)	BOSS0235
001643	F77=(E77-E75*F57)	BOSS0236
001646	F78=(E78-E75*F58)	BOSS0237
001651	F79=(E79-E75*F59)	BOSS0238
001654	F86=(E86-E85*F56)	BOSS0239
001657	F87=(E87-E85*F57)	BOSS0240
001662	F88=(E88-E85*F58)	BOSS0241
001665	F89=(E89-E85*E59/E55)	BOSS0242
001670	G67=F67/F66	BOSS0243
001672	G68=F68/F66	BOSS0244
001674	G69=F69/F66	BOSS0245
001675	G17=(F17-F16*G67)	BOSS0246
001700	G18=(F18-F16*G68)	BOSS0247
001703	G19=(F19-F16*G69)	BOSS0248
001706	G27=(F27-F26*G67)	BOSS0249
001711	G28=(F28-F26*G68)	BOSS0250
001714	G29=(F29-F26*G69)	BOSS0251

001717	G37=(F37-F36*G67)	BOSS0252
001727	G38=(F38-F36*G68)	BOSS0253
001725	G39=(F39-F36*G69)	BOSS0254
001727	G47=(F47-F46*G67)	BOSS0255
001732	G48=(F48-F46*G68)	BOSS0256
001735	G49=(F49-F46*G69)	BOSS0257
001740	G57=(F57-F56*G67)	BOSS0258
001743	G58=(F58-F56*G68)	BOSS0259
001746	G59=(F59-F56*G69)	BOSS0260
001750	G77=(F77-F76*G67)	BOSS0261
001753	G78=(F78-F76*G68)	BOSS0262
001756	G79=(F79-F76*G69)	BOSS0263
001761	G87=(F87-F86*G67)	BOSS0264
001764	G88=(F88-F86*G68)	BOSS0265
001767	G89=(F89-F86*G69)	BOSS0266
001771	H78=G78/G77	BOSS0267
001773	H79=G79/G77	BOSS0268
001775	H18=(G18-G17*H78)	BOSS0269
002000	H19=(G19-G17*H79)	BOSS0270
002003	H28=(G28-G27*H78)	BOSS0271
002006	H29=(G29-G27*H79)	BOSS0272
002010	H38=(G38-G37*H78)	BOSS0273
002013	H39=(G39-G37*H79)	BOSS0274
002016	H48=(G48-G47*H78)	BOSS0275
002021	H49=(G49-G47*H79)	BOSS0276
002023	H58=(G58-G57*H78)	BOSS0277
002026	H59=(G59-G57*H79)	BOSS0278
002031	H68=(G68-G67*H78)	BOSS0279
002034	H69=(G69-G67*H79)	BOSS0280
002036	H88=(G88-G87*H78)	BOSS0281
002041	H89=(G89-G87*H79)	BOSS0282
002044	089=H89/H88	BOSS0283
002046	0(1) =(H19-H18*089)	BOSS0284
002051	0(2) =(H29-H28*089)	BOSS0285
002056	0(3) =(H39-H38*089)	BOSS0286
002060	0(4) =(H49-H48*089)	BOSS0287
002063	0(5) =(H59-H58*089)	BOSS0288
002067	0(6) =(H69-H68*089)	BOSS0289
002072	0(7) =(H79-H78*089)	BOSS0290
002076	0(8) =089	BOSS0291
002077	RETURN	BOSS0292
002077	END	BOSS0293

C
C
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000002
000003

RETURN
END

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SUBROUTINE PLOTS

```
C
C*****
C#
C# PURPOSE
C#
C# SUBROUTINE PLOTS GIVES PLOTS OF THE SOLUTIONS OF THE EQUATIONS
C# OF MOTION, I.E., PLOTS OF THE ANGLES AND ANGULAR VELOCITIES AS A
C# FUNCTION OF TIME. PLOTS OF THETA VS PHI ARE ALSO PROVIDED.
C#
C# PROPRIETARY
C#
C# NONE
C#
C# USE
C#
C# CALL PLOTS
C#
C# PARAMETERS
C#
C# NONE
C#
C# AUTHOR/IMPLEMENTOR
C#
C# D.R. DORSEY, JR - B. SEAY
C#
C# DATE RELEASED
C#
C# SEPTEMBER 14, 1973
C#
C# LATEST REVISION
C#
C# SEPTEMBER 14, 1973
C#
C*****
C
C THIS IS A DUMMY SUBROUTINE TO SATISFY THE LOADER.
C IT HAS BEEN DECIDED THAT THE PLOTTING ROUTINES FOR PROGRAM
C LACATE CAN BEST BE HANDLED AS POSTPROCESSOR ROUTINES RUN
C SEPARATELY FROM LACATE. FOR PLOTS OF THE SOLUTIONS OF THE
C EQUATIONS OF MOTION, REFER TO PROGRAM AVPLOT. FOR THE TWO-
C DIMENSIONAL THETA-PHI PLOTS, REFER TO PROGRAM TPLOT.
```

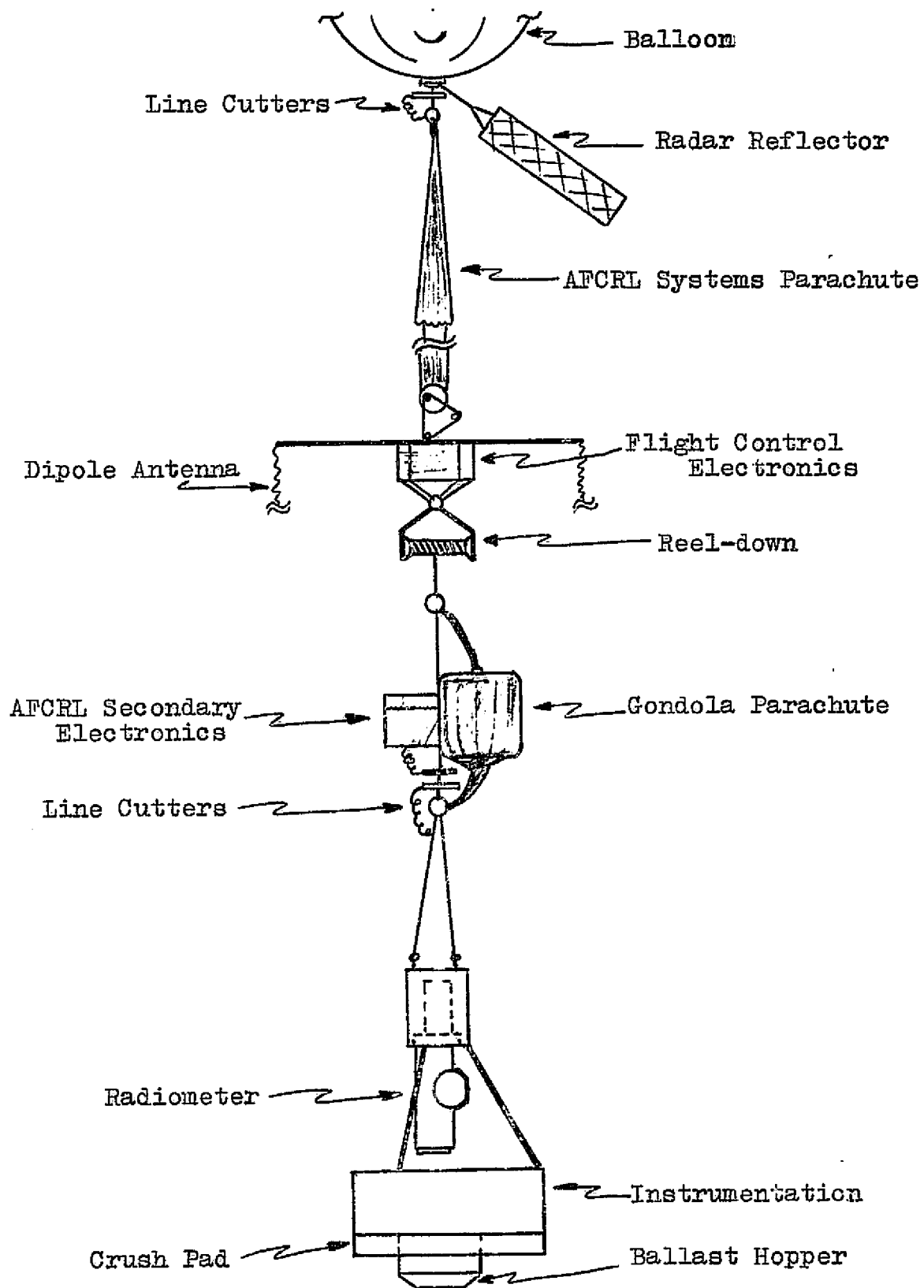


Figure 1a. IACATE Balloon System - General Structure

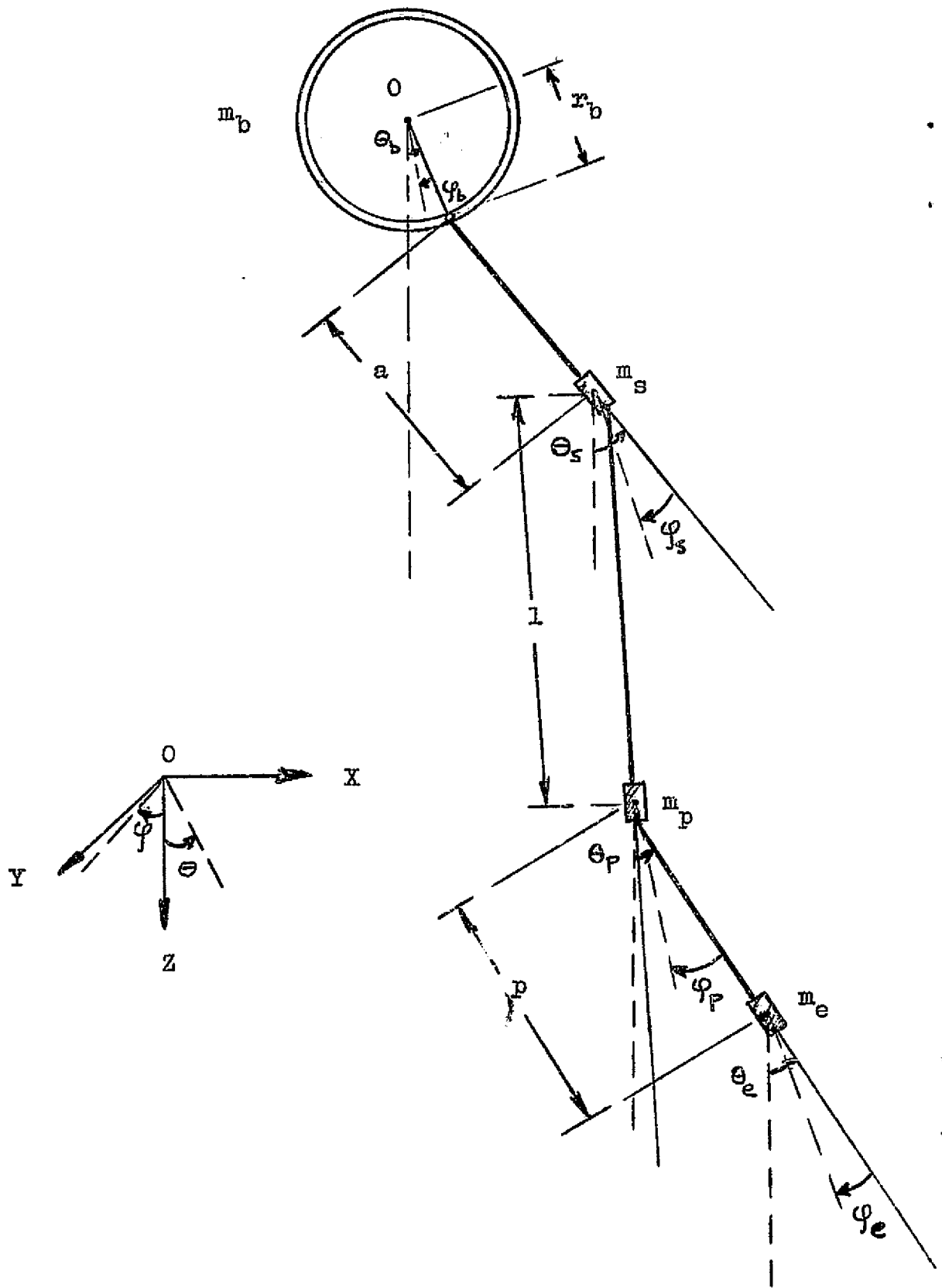


Figure 1b. IACARE Balloon Idealized Model

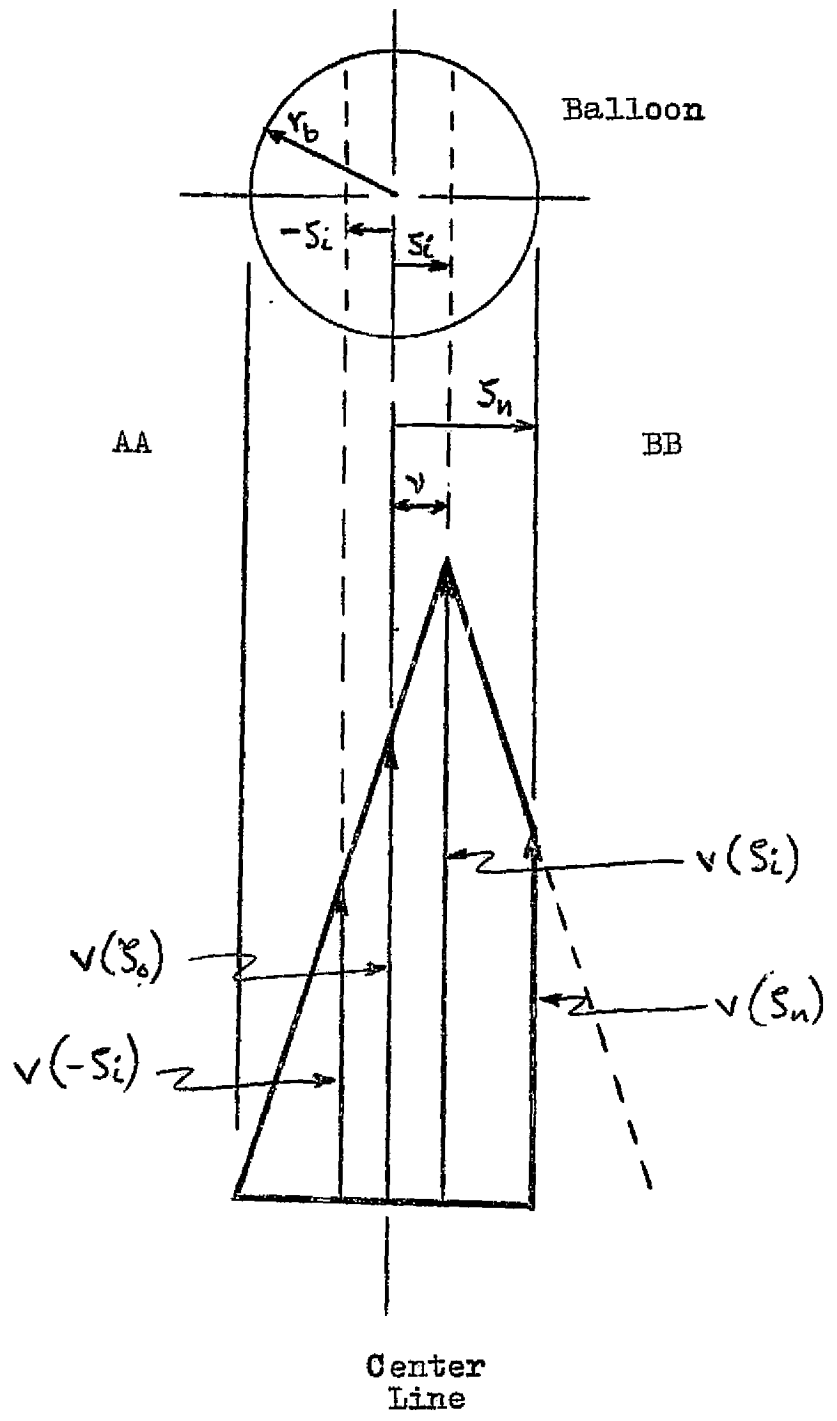


Figure 2. Geometry Definitions For Gust Model

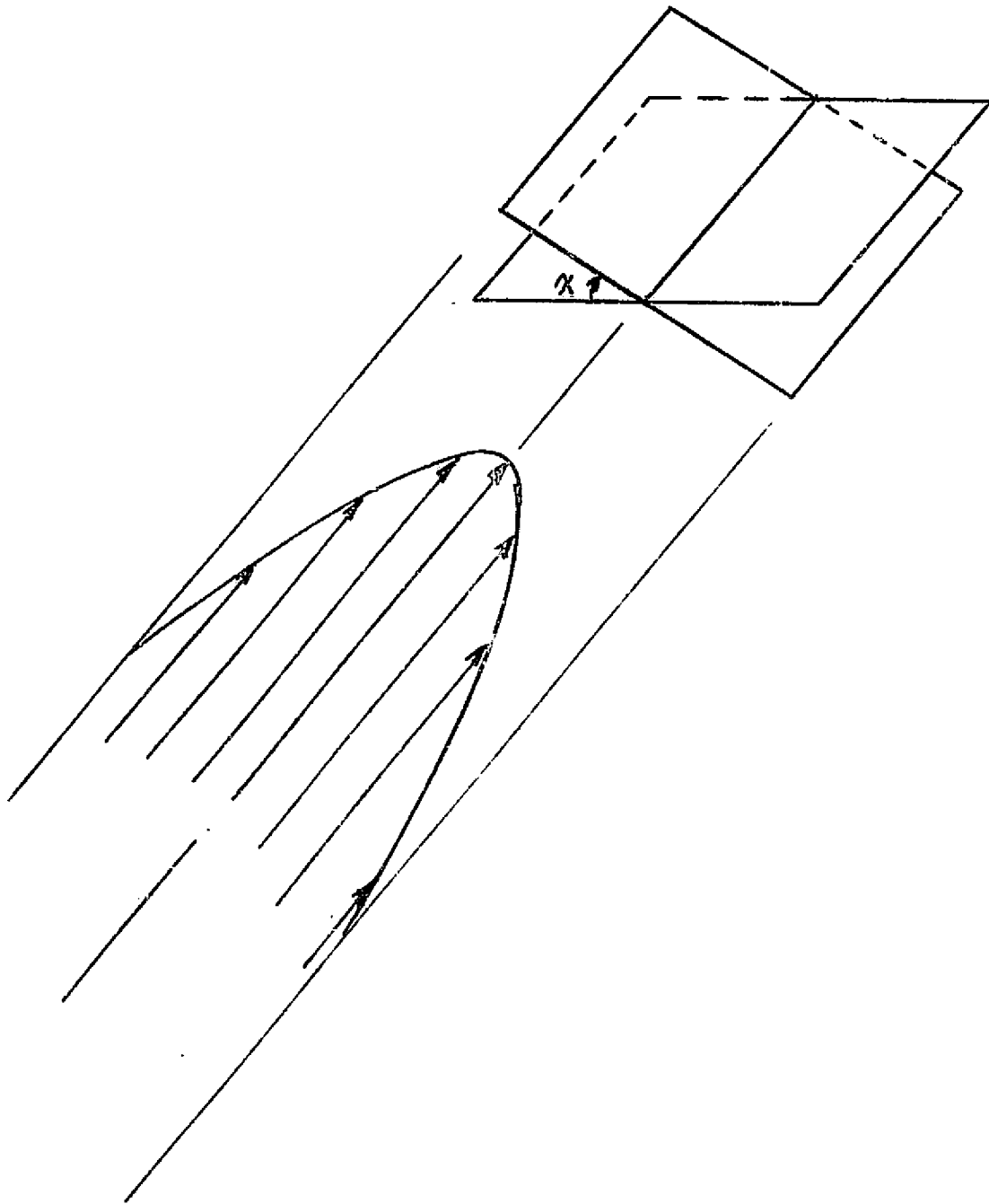


Figure 3. Planar Gust Profile Rotated about Azimuth Plane

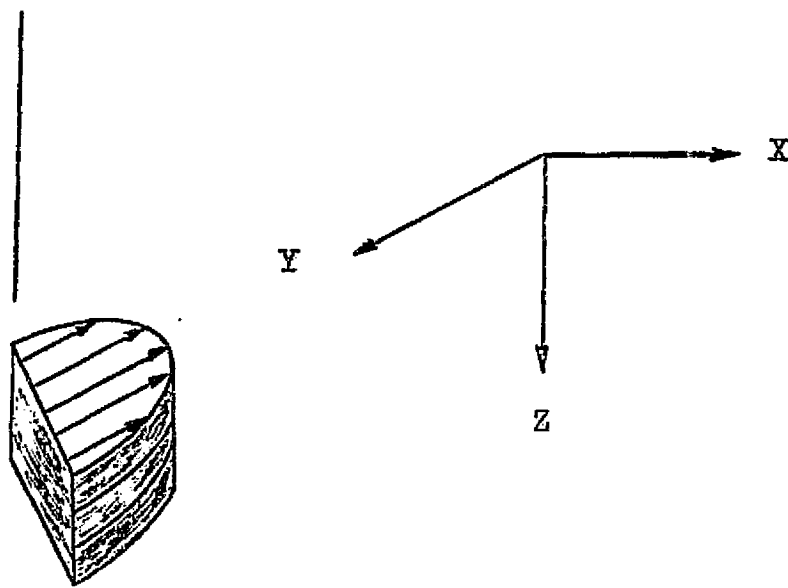


Figure 4. Planar Stacked Gust Profile Parallel to Azimuth Plane

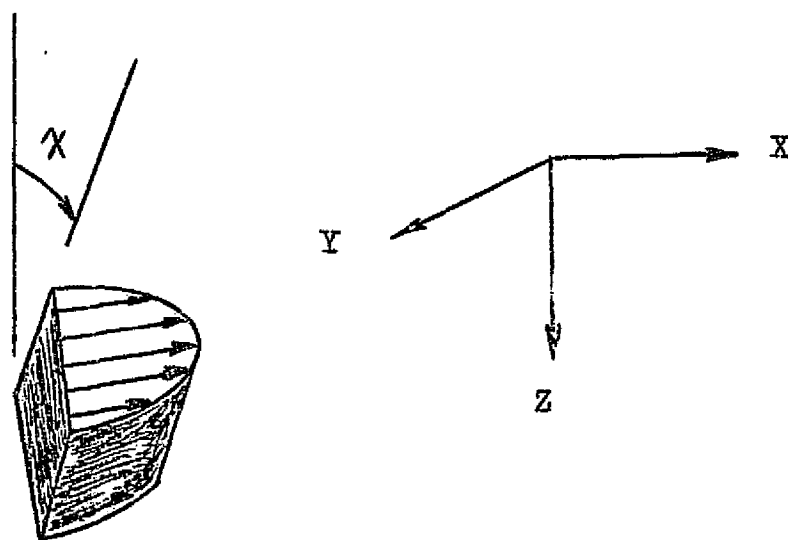


Figure 5. Planar Stacked Gust Profile Rotated about Azimuth Plane

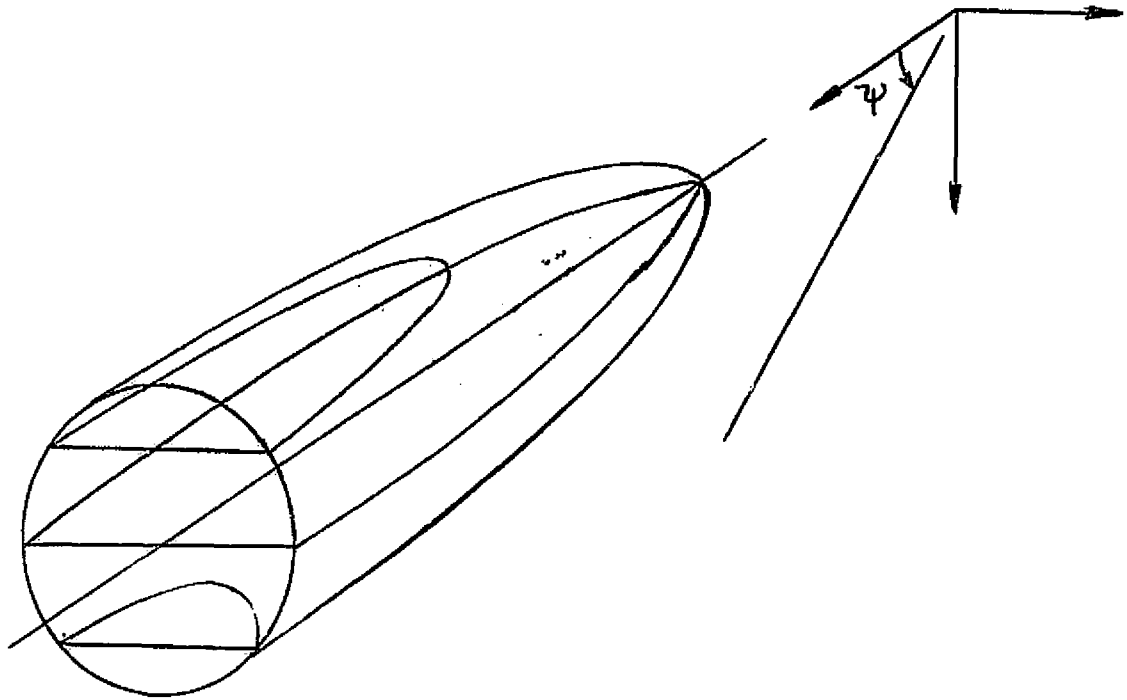


Figure 6. General Three-Dimensional Gust Profile
In Form of a Body of Revolution

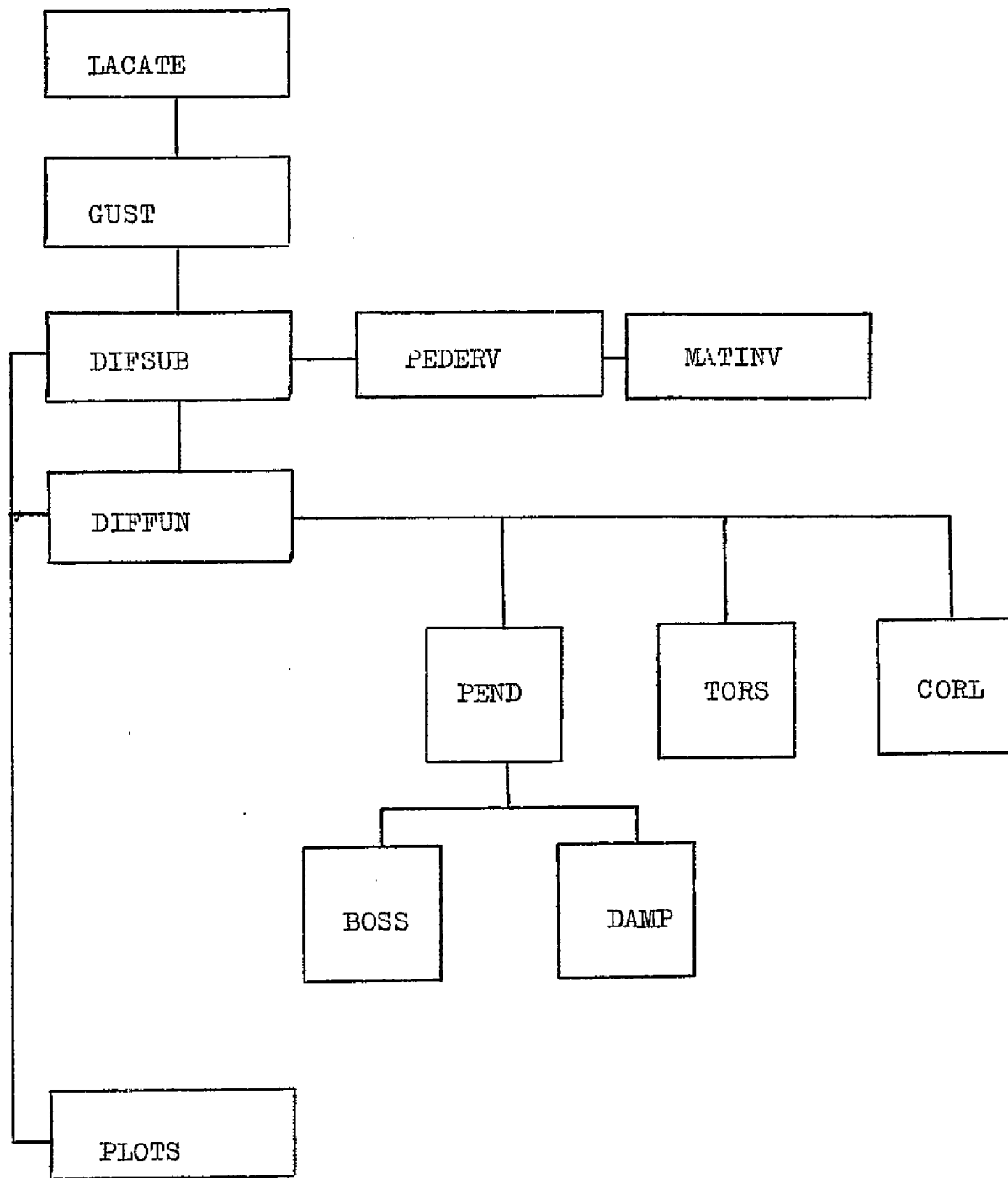


Figure 7. LACATE Program Structure