

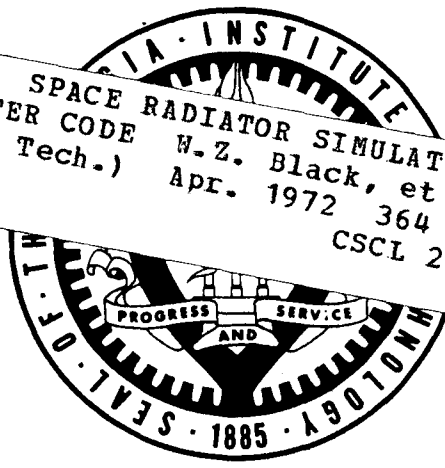
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GEORGIA INSTITUTE OF TECHNOLOGY  
School of Mechanical Engineering  
Atlanta, Georgia

SPACE RADIATOR SIMULATION  
MANUAL FOR COMPUTER CODE

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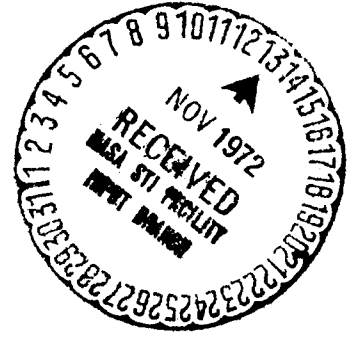
Contract No. NAS 9-10415

by

William Z. Black and Wolfgang Wulff

Sponsored by the  
Power Generation Branch  
Manned Spacecraft Center  
NATIONAL AERONAUTICS AND SPACE ADMINISTRATION  
Houston, Texas

April 1972



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SCHOOL OF MECHANICAL ENGINEERING  
ATLANTA, GEORGIA

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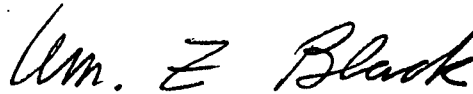
William Z. Black and Wolfgang Wulff

SCHOOL OF MECHANICAL ENGINEERING  
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ATLANTA, GEORGIA 30332

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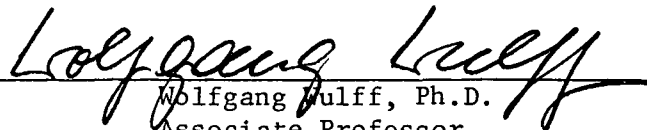
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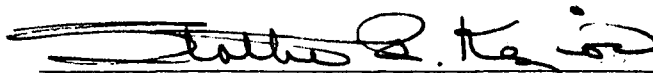
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## FOREWORD

This report constitutes the Users Manual that accompanies the computer simulation of a space radiator system. It summarizes a two year project carried out by the School of Mechanical Engineering at the Georgia Institute of Technology, Atlanta, Georgia for the NASA Manned Spacecraft Center, Houston, Texas under Contract No. NAS 9-10415. The results of the analysis phase of the program are contained in a separate report entitled "Space Radiator Simulation - System Analysis". A third report submitted under this contract covers the development of a simplified system simulation and the initial phases of a system optimization procedure.

The contract was entitled "Study of Design Parameters of Space Base and Space Shuttle Heat Rejection Systems". The work was monitored by Dr. W. E. Simon of the Power Generation Branch of NASA MSC, Houston, Texas, and was carried out by Dr. W. Z. Black and Dr. W. Wulff as Co-Investigators and Mr. S. M. Morcos, Mr. S. L. Yao and Mr. R. M. Hinson graduate students in the School of Mechanical Engineering under the direction of Dr. S. P. Kezios. Mr. C. R. Caldarale of the Rich Electronic Computing Center also contributed to this report.

The Work carried out by Dr. W. Z. Black is reflected in Section B.2 a and b and Section B.3, d through f, of the Permanent Part of the program. The program units of the Selective part described in Section B.3 were completed by Mr. Morcos under the direction of the Co-Investigators.

Dr. Black is responsible for the coding of programs No. 14 through 22, for the supervision of coding programs No. 23 through 30 which was carried out by Mr. Caldarale and Mr. Hinson and for the supervision of coding programs No. 51 through 60 which was carried out by Mr. Morcos. Dr. W. Wulff is responsible for the general program structure

described in Section B.1 and the program units described in the Sections B.3a through c and B.3g of the Permanent Part of the program. Dr. Wulff coded programs 1 through 7, 31 through 38 and 40 through 42 and supervised Mr. Yao in the coding of programs No. 8 through 13 and No. 39 and supervised Mr. Morcos in the coding of programs No. 43 through 50 and No. 61 and 62.

## SUMMARY

This manual describes a computer program that simulates the performance of a space radiator. The program basically consists of a rigorous analysis which analyzes a symmetrical fin panel and an approximate analysis that predicts system characteristics for cases of non-symmetrical operation. The rigorous analysis accounts for both transient and steady state performance including aerodynamic and radiant heating of the radiator system. The approximate analysis considers only steady state operation with no aerodynamic heating.

The first section of this manual contains a description of the radiator system and instructions to the user for program operation. The input required for the execution of all program options is described. Several examples of program output are contained in this section. Sample output includes the radiator performance during ascent, reentry and orbit.

The second section of the user's manual describes the structure of the entire program as well as the function of each program unit. All interfaces between the various program units are described so that the user may remove a single unit or group of program units that have general utility for use in other analyses. This section concludes with a program listing.

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## A. USERS INSTRUCTIONS

### 1. Program Objectives and Capabilities

The program units described in this manual constitute a transient heat transfer analysis of a space radiator heat rejection system. The analysis is intended to simulate the radiator system of the space base and the space shuttle for conditions of arbitrarily prescribed combinations of aerodynamic heating and incident solar, earth albedo and planetary irradiation.

An exact analysis is performed for the radiator system consisting of equally spaced parallel coolant channels existing in one plane connected by plane fin panels of trapezoidal cross-section. The midplane between tubes and a parallel plane passing through the tube axis are assumed to be planes of symmetry for the purpose of the exact analysis. The program results are extended over the entire radiator panel on the basis of this symmetry.

An approximate analysis is used to extend the exact program simulation to cover the situation of the non-symmetrical radiator panel. This extension predicts on the assumption of one-dimensional analysis, the system performance when the radiator panel is curved and when the tubes are unequally spaced, non-symmetrically loaded or U-shaped.

The exact system analysis predicts both transient and steady-state, two-dimensional temperature profiles, local and total heat transfer rates and coolant fluid temperature and pressure profiles in the flow channel. The analysis also predicts the protection layer thickness which covers the coolant channel and the inlet and exit manifold tubing. Total system mass and projected area are calculated.

The program units described in this manual serve to simulate the radiator system performance. Even though successive runs with system parameter variations are feasible, the task of system optimization is dealt with in the simplified analysis which is described in detail in the Simplified Analysis Manual.

The purpose of this manual is to describe the program units that make up the computer simulation, the necessary input parameters for program execution

and the typical program output for simulated radiator operation. The deck composition is described in part 2, and the input data set is explained in part 3. Part 4 is devoted to a description of the program output and part 5 gives examples of typical radiator performance for ascent, reentry and orbital conditions. Section B describes each subprogram in detail by describing their input and output variables as well as the function of each program unit.

The simulation program is separated into modular units or subprograms so that units of general utility may be removed and run separately. This arrangement also facilitates program modifications if future changes are required. The subprogram concept also reduces the extent of program checking when new program units are added to the existing program package.

The entire program consists of 62 separate program units and the Data Set. The source language is FORTRAN V. The program is executed from catalogued files, using temporary data files as batch processing on the UNIVAC 1108 computer operating under EXEC 8, Version 27.20.143, at the Georgia Tech Rich Electronic Computer Center.

Except for the control statements necessary to collect the program into elements to be stored on mass storage for subsequent program execution from mass storage, the Executive Control Statements are not discussed. Control statements may be freely chosen by the user in accordance with the Executive Control Language. All statements are in 026 key punch code.

#### a. System Description

The exact simulation program predicts the performance characteristics of the typical coolant channel and fin segment shown in Fig. 1. The fin profile has a trapezoidal cross-sectional area. The parallel plane passing through the tube axis and the plane bisecting two adjacent tubes are assumed to be planes of symmetry.

The program has property options that allow the user to select from a number of different coolant fluids, structural and meteoroid protection materials and surface coatings. The materials available for the protection layer, tube and fin are beryllium, copper or aluminum in any combination. The coolant fluids may be selected from a gas, several liquids or a liquid metal.

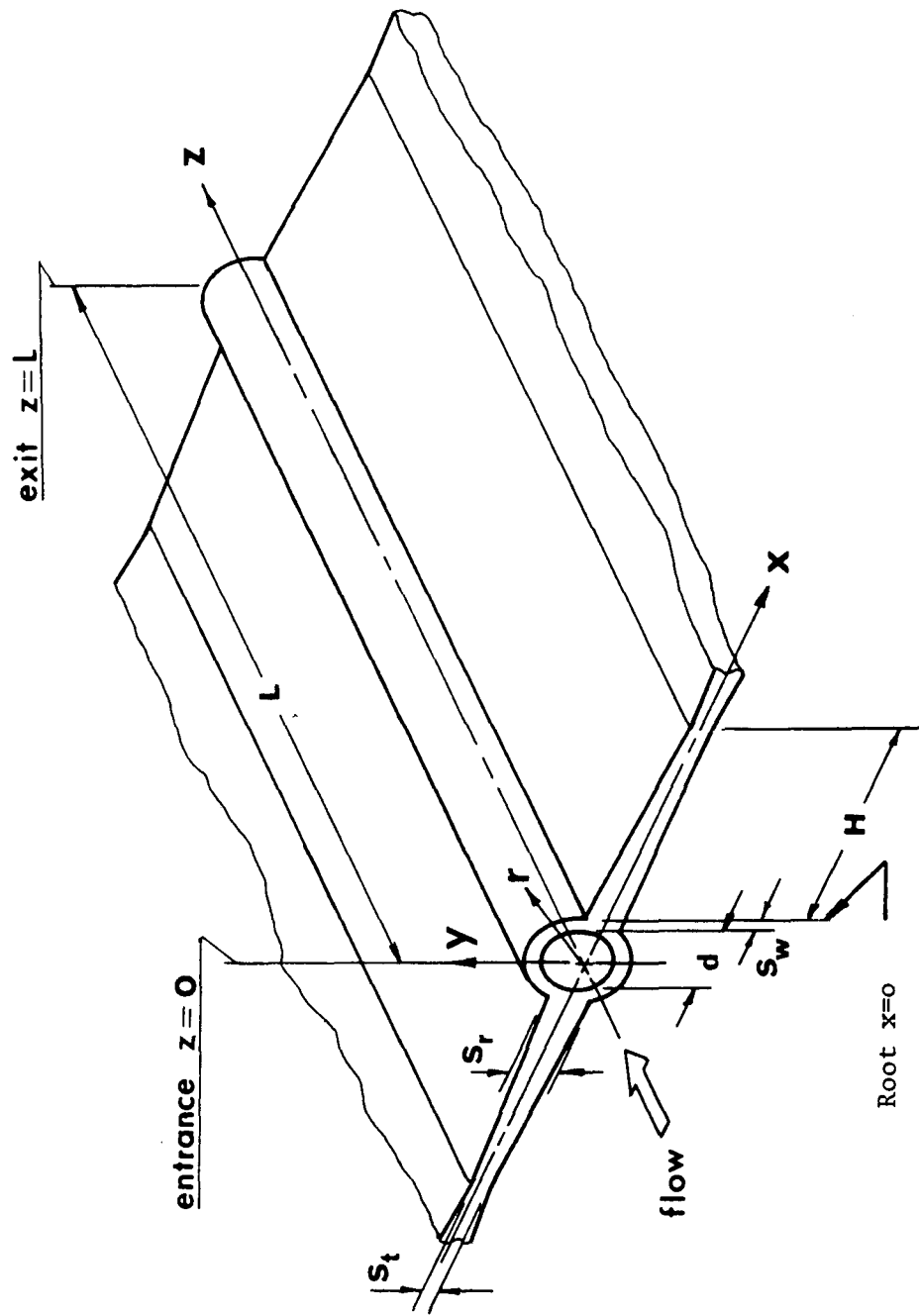


Fig. 1 Typical Fin Element for Exact Analysis

Possible choices are helium, silicon oil, the two fluorochemicals FC-43 and FC-75 and the liquid metal NaK. The entire radiator system is covered with the same passive thermal coating Z-93 which has optically diffuse but wavelength and temperature dependent optical properties.

The approximate analysis predicts the performance of the non-symmetrical radiator panels shown in Fig. 2. This one-dimensional analysis predicts the approximate location of the adiabatic plane separating two adjacent coolant channels when the radiator panel is curved (Fig. 2a) or the coolant channels are shaped in form of the letter U (Fig. 2b). Other situations which the approximate analysis is capable of treating involve the radiator panel with non-uniformly spaced coolant channels and coolant channels that do not all have identical inlet fluid mass flow rates or identical inlet fluid temperatures.

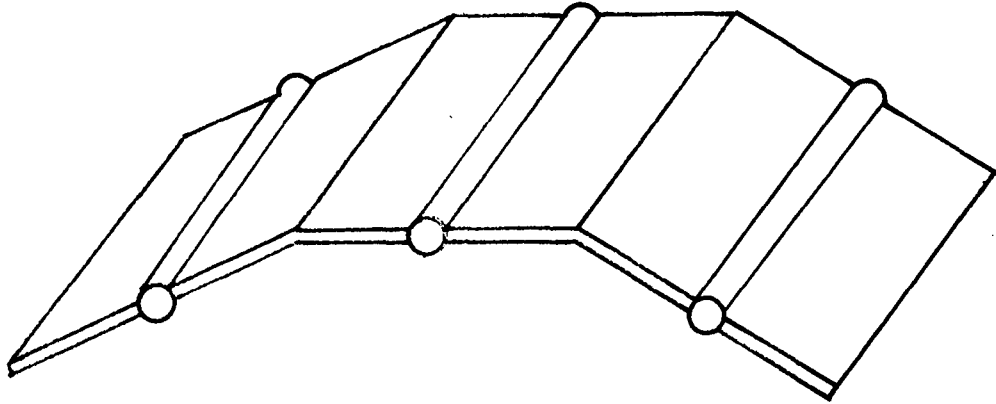
#### b. Operating Conditions

The program simulates radiator characteristics for ascent and reentry of the shuttle radiator system as well as orbital conditions for the shuttle and space base radiator. Typical system performance for these conditions are shown in part 5 of this manual.

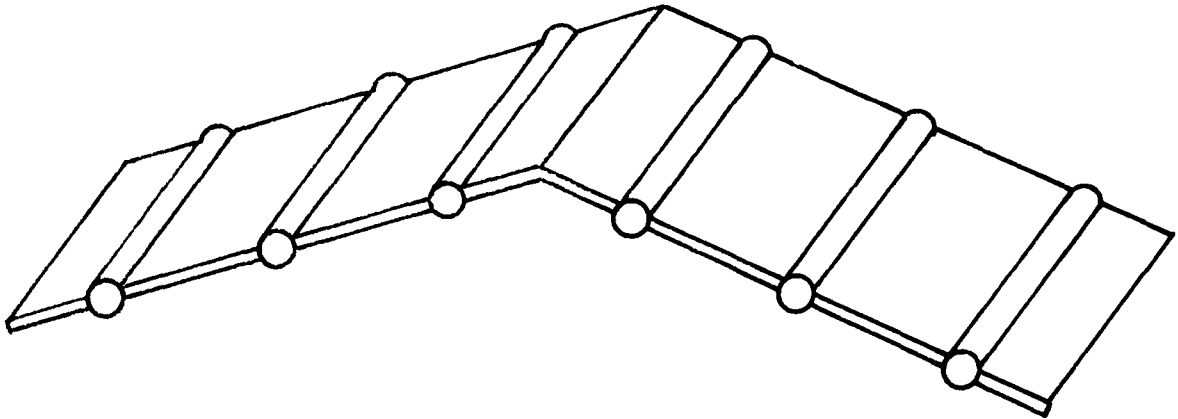
The altitude and velocity profiles of the shuttle vehicle during ascent and reentry are supplied by the user. These profiles are read into the program from card images.

The irradiation data to be supplied by the user is subdivided into three basic categories: solar, earth albedo and planetary irradiation. This data may be either read from card images or from a previously prepared data file which contains the irradiation data in the format specified by the MRI radiant flux program.

The program is capable of system simulation for both transient and steady state conditions. Under the steady state option, integration of the system differential equations proceeds from the user supplied initial condition until steady state is reached. All environmental conditions such as incident radiant flux remain constant during the integration process. Under the transient option, the environmental conditions are allowed to change with time as the integration process proceeds.



(a) Curved Radiator Panel



(b) U Shaped Coolant Channels

Fig. 2 Radiator Panel For Approximate Analysis

Provision has been made within the program to allow for the variation in coolant fluid temperature and mass flow at the inlet plane of the coolant channel. These variations must be supplied by the user as sets of discrete data values punched on cards.

## 2. The Deck Composition

The program structure is dictated by the calling sequence of the Runge-Kutta-Simpson integration procedure (see Section III-2 of the System Analysis Manual). The Source Deck consists of two parts. The first part, referred to as the Permanent Part, requires no attention from the user unless he plans major program modifications, for instance to accommodate geometrical configurations other than the one considered here. The Permanent Part contains the first 42 program units.

The second part, called the Selective Part, is assembled from case to case by the user in accordance with his choice of the

- (i) coolant fluid
- (ii) flow channel material
- (iii) meteoroid protection layer material
- (iv) fin panel material
- (v) thermal control coating

There are always 20 program units in the Selective Part.

Following the Source Deck is the Data Set which is discussed in Section A-3. It consists of six groups, one for each of the following input specifications:

- (i) parameters which specify non-symmetrical radiator panel
- (ii) ascent and reentry profiles
- (iii) solar, albedo and planetary irradiation
- (iv) coolant fluid inlet temperature and mass flux histories
- (v) system specifications
- (vi) program options.

Most of the user's activities will be reflected in his manipulation of the Data Set, especially of the data in groups (v) and (vi) above.



### 3. The Input Data Set

Input data are divided into six groups and assembled in the order of increasing frequency of changes so that data records most likely to be changed are at the end of the Data Set. This enables the user to keep large portions of the Job Deck on mass storage. The six groups in the Data Set define

- (i) the parameters which specify the non-symmetrical radiator panel
- (ii) the ascent and reentry profiles
- (iii) the incident thermal radiant flux as a function of time
- (iv) the coolant fluid inlet temperature and mass flux histories
- (v) the system specifications, other than material selections
- (vi) the options of program execution.

The program may be executed successively after reading new data records beginning at any of the six data groups listed above. Provisions have been also made for restarting the program by reading new data records from one group and skipping the following data group or only skipping portions of the succeeding data groups. The restart options are determined by the value of the restart integer MRSTRT. A description of this integer is given in part (f) of this section.

None of the data groups contains a fixed number of records. The number of records in groups (ii), (iii) and (iv) are determined and specified to the computer by the user. Groups (i), (v) and (vi) contain a minimum of records as specified below; the number of additional records in these groups and of additional sets of groups (ii), (iii) and (iv) depend on the number of additional executions and parameter variations during the same job. In the description of data records (card images) the numbering starts from one in each group.

a) Alphabetical List of Input Data

All input variables needed for program operation are listed below with their units and a brief description of the meaning of the variable.

VARIABLE NAME	UNITS	REFERENCED IN	DESCRIPTION
ABEMF		FINA	ratio of solar absorptivity to infrared emissivity of coating material
AL	ft	TUBEA	array of tube lengths
ALIMIT		RUNØPT	absolute error limit per integration step
ALPHA	gm/(day ft <sup>2</sup> )	PRØTLR	constant relating meteoroid flux to mass
ALTA	ft	ascent and reentry data	elements in altitude array used for ascent altitude profile of shuttle
ALTR	ft	ascent and reentry data	elements in altitude array used for reentry altitude profile of shuttle
AMAN	ft <sup>2</sup>	MANIFD	total manifold area projected into plane of fin
AN		PROTLR	see equation 8.6 System Analysis Manual
ATK		PRØTLR	see equation 8.8 in System Analysis Manual
BTA		PRØTLR	constant relating meteoroid flux to mass
D	in	TUBEA	tube internal diameter
DITBI	in	TUBE	internal tube diameter
DTWRTE	hr	RUNØPT	fixed time interval between data printout
GAMMA		PRØTLR	constant used to adjust observed to predicted meteoroid penetration depth

H	in	FINA	array of fin heights
HFNI	in	FIN	fin height from root to tip
ICASE		QNML,MRI	integer value which specifies set of irradiation values on MRI file
ISYM		SYSTEM	integer which specifies type of non-symmetrical loading of tubes
ITAPE		QNML,MRI	integer value to specify card input (= 0) or MRI file input ( $\neq$ 0) for irradiation data
LFLD		RUNOPT	control integer used to select between variable or fixed inlet fluid properties
LIMWRT		RUNOPT	maximum number of data recordings during integration toward steady state
LTS		RUNOPT	control integer used to select initial system temperature
LTT		RUNOPT	control integer used to select between temperature dependent or independent surface coating properties
M	lbm/hr	FLUIDA	array of mass flow rates at inlets to tubes
MDOTI	lbm/hr	FLOW	total coolant mass flow rate
MRSTRT		GINPT	control integer used for selection of reentry point for restarting program
MSTOTR		RUNOPT	control integer for selecting transient or steady state computation
MZ		TUBE	number of nodal points along tube axis ( $\leq$ 10)
NA		ascent and reentry data	number of instances for specified ascent velocity and altitude data
NCNV		RUNOPT	control integer for selection of orbit, ascent or reentry

NFLD		inlet coolant fluid history	number of mass flow rate and temperature data values
NR		ascent and reentry data	number of instances for specified reentry velocity and altitude data
NRMP		PRØTLR	number of radial nodal points in the protection layer ( $\leq 5$ )
NRTBI		TUBE	number of radial nodal points in tube wall ( $\leq 5$ )
NSRD		SYSTEM	number of sides of the radiator panel which radiate (1 or 2)
NT		SYSTEM	number of flat symmetrically loaded panels
NTBS		TUBE	number of flow channels
NTM		QNML,MRI	number of instances for specified irradiation data ( $< 100$ )
NX		FIN	number of nodal points on fin perpendicular to flow channel ( $\leq 10$ )
PHI		PROTLR	see equation 8.1 System Analysis Manual
PHIN	degrees	MRI	array of angles that relate fin panel orientation with MRI data
PHN	degrees	QNML	angle of panel relative to MRI data
PIN	lbf/ft <sup>2</sup>	FLUIDA	array of pressures at inlet to tubes
PO	lbf/ft <sup>2</sup>	FLØW	entrance pressure into flow channel
PRØB		PRØTLR	probability of no damage caused by meteoroid impact
RHØFNI	lbm/ft <sup>3</sup>	FIN	fin material density
RHØMET	g/cm <sup>3</sup>	PRØTLR	meteoroid density
RHØMPI	lbm/ft <sup>3</sup>	PRØTLR	protection layer density

RHØTBI	lbm/ft <sup>3</sup>	TUBE	tube wall density
RLIMIT		RUNØPT	relative error per integration step
SRØØTI	in	FIN	fin root thickness
STAGX	ft	FIN	distance from stagnation point on shuttle to center of radiator panel along a streamline
STBI	in	TUBE	tube wall thickness
STIPI	in	FIN	fin tip thickness
TA	sec	ascent and reentry data	elements in time array used for ascent velocity and elevation profiles
TAU	days	PRØTLR	time radiator is exposed to meteoroid environment
TEND	hr	RUNØPT	termination time for transient performance calculation
TH	in	FINA	array of fin thickness
THETA		PRØTLR	see equation 8.1 System Analysis Manual
TI	R	RUNØPT	initial system temperature
TIFLD	R	coolant fluid inlet temperature history	array of coolant fluid inlet temperatures
TIN	R	FLUIDA	array of temperatures at inlets to tubes
TO	R	FLØW	entrance temperature into flow channel
TØ	min	QNML,MRI	time in minutes at which irradiation data is required
TR	sec	ascent and reentry data	elements in time array used for reentry velocity and elevation profiles
TSTAR	R	SYSTEM	array of sink temperatures

VELA	ft/sec	ascent and reentry data	elements in velocity array used for ascent velocity profile of shuttle
VELM	ft/sec	PRØTLR	average meteoroid velocity
VELR	ft/sec	ascent and reentry data	elements in velocity array used for reentry velocity profile of shuttle
VERTX	ft	FIN	total dimension of radiator panel in direction parallel to acceleration of gravity
W1	in	TUBEA	array of distances between side segments of tubes when U-shaped
W2	in	TUBEA	array of distances between base segments of tubes when U-shaped
WIFLD	lbm/hr	coolant fluid inlet mass flow rate history	array of coolant fluid inlet mass flow rates
XL	ft	TUBE	tube length

b) Parameters Specifying Non-Symmetrical Radiator Panels

Normal program operation assumes that radiator conditions are symmetrical about each coolant fluid channel. Several factors may destroy this symmetry. One may be non-uniform spacing of the tubes. Another factor may be different inlet temperatures or mass flow rates into adjacent tubes. A third factor could be attributed to a curved panel resulting in a variation to sink temperature between adjacent fin segments. These factors are specified in the NAMELIST's that follow.

Numerical values for a non-symmetrical panel, for irradiation data and for system specifications are read into the computer in NAMELIST format, one NAMELIST each with name and input data as follows:

i) for non-symmetrical panel specifications (see Section (b))

SYSTEM for selection of system parameters when panels and/or flow conditions are non-symmetrical

TUBEA for non-symmetrical tube specifications

FINA for non-symmetrical fin specifications

FLUIDA for non-symmetrical fluid specifications

MRI for non-symmetrical irradiation data

ii) for irradiation specifications (see Section (c))

QNML for irradiation data under symmetrical radiator panel conditions

iii) for system specifications (see Section (f))

TUBE for flow channel specifications

FLØW for coolant fluid flow specifications

FIN for fin parameter specifications

PRØTLR for the protection layer specifications

MANIFD for the manifold input data.

Each input data record under NAMELIST format has

- (i) a blank in its first column,
- (ii) a \$ - sign in its second column,
- (iii) the NAMELIST name, starting in the third column, followed by one or more blanks,

(iv) the string of data in free field format as follows:

$$A = b ,$$

where (A) is the variable name and (b) is a number; imbedded blanks between (A), (=), (b), ( , ) and a following (A) are permissible,

(v) the comma of the last datum is to be placed by a \$ - sign which delimits the NAMELIST data string.

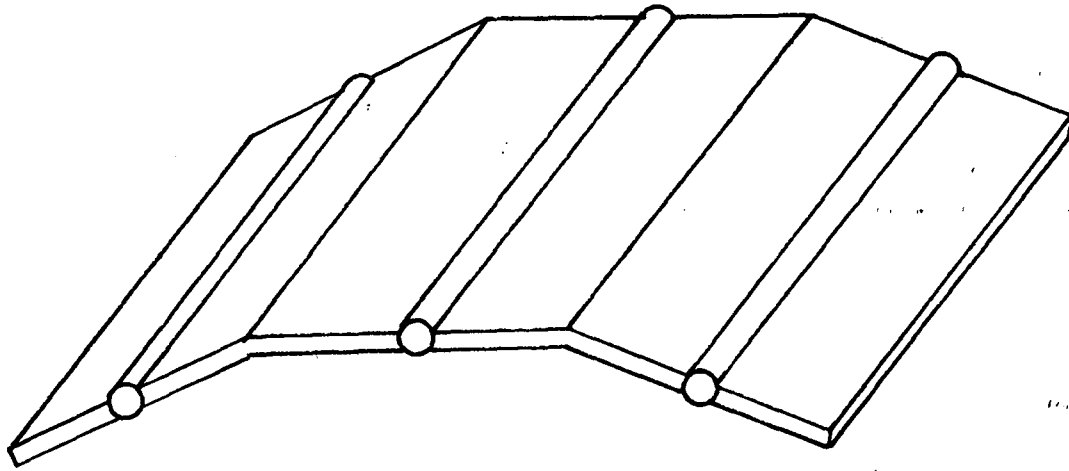
For more details concerning NAMELIST formatting, the reader may consult standard texts on FORTRAN PROGRAMMING. The NAMELIST data strings must be sequenced in the order in which they are listed above.

i) NAMELIST/SYSTEM/contains 4 variables

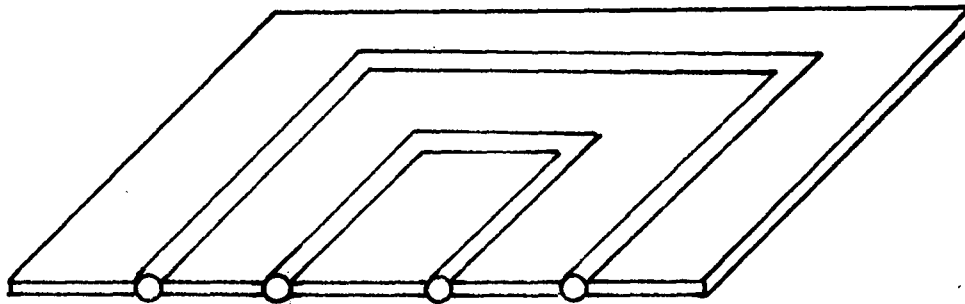
NT	number of flat symmetrically loaded panels, each panel containing NTBS symmetrically loaded tubes (see Fig.3 )
NSRD	= 1 for one side of the panel radiating = 2 for both sides of the panel radiating Note: if panel is curved NSRD is set equal to 1
ISYM	= 0 for symmetrical panel in one plane and straight parallel equally spaced tubes = 1 for non-symmetrical curved panel consisting of straight fin segments and straight parallel tubes = 2 for non-symmetrical panel in one plane with U-shaped tubes = 3 for non-symmetrical curved panel consisting of straight fin segments and U-shaped tubes = 4 for non-symmetrical panel in one plane with straight parallel tubes but non-symmetrical loading
TSTAR	an (NT,3) array of sink temperatures in °R. These values are ignored if MRI file is used for incident flux data (see Section (d)). Also: If the radiator panel is flat only (1, NSRD) of the array is required.

The following four NAMELIST's are not read and should not be placed in the program if ISYM = 0. If ISYM ≠ 0 these four NAMELIST's should appear in the order shown.





NT = 3  
NTBS = 1



NT = 2  
NTBS = 3

Fig. 3

Examples of the Input Parameters NT and NTBS

ii) NAMELIST/TUBEA/contains 4 variables

- D            the single value of all tube internal diameters in inches
- AL           an array of NT tube lengths in feet. These values are used only for straight tubes. For curved tubes, the lengths are calculated from the values of W1 and W2 which are read in below
- W1           distance between U-shaped tubes along inlet and exit portion of the tube in inches (i.e., both side segments of the U) NT + 1 values
- W2           distance between U-shaped tubes along bottom segment of U in inches, NT + 1 values (see Fig. 4 for details of dimensions W1 and W2)

iii) NAMELIST/FINA/contains 3 variables

- H            NT + 1 values of the fin height in inches from tube centerline to midplane between tubes. This value is used on for the case of straight tubes. Fin height for cases of the U tube is calculated from W1 and W2 in NAMELIST TUBEA
- TH           NT values for the fin thickness in inches
- ABEMF       ratio of the solar absorptivity to infrared emissivity of the surface of the radiator system

iv) NAMELIST/FLUIDA/contains 3 variables

- TIN          NT values for the inlet fluid temperatures in R
- PIN          NT values for the inlet fluid pressure in lbf/ft<sup>2</sup>
- M            NT values for the inlet fluid mass flow rate in lbm/hr

v) NAMELIST/MRI/contains 5 variables

- ITAPE = 0    to read irradiation data from cards
- > 0    to read irradiation data from MRI file. Value of ITAPE designates the input unit used to read file data
- ICASE       integer used to designate the case number of stored data to be used for input. ICASE = 3, for example, would designate the third set of data stored on the MRI file to be used for current input. If ITAPE = 0, this integer is ignored.

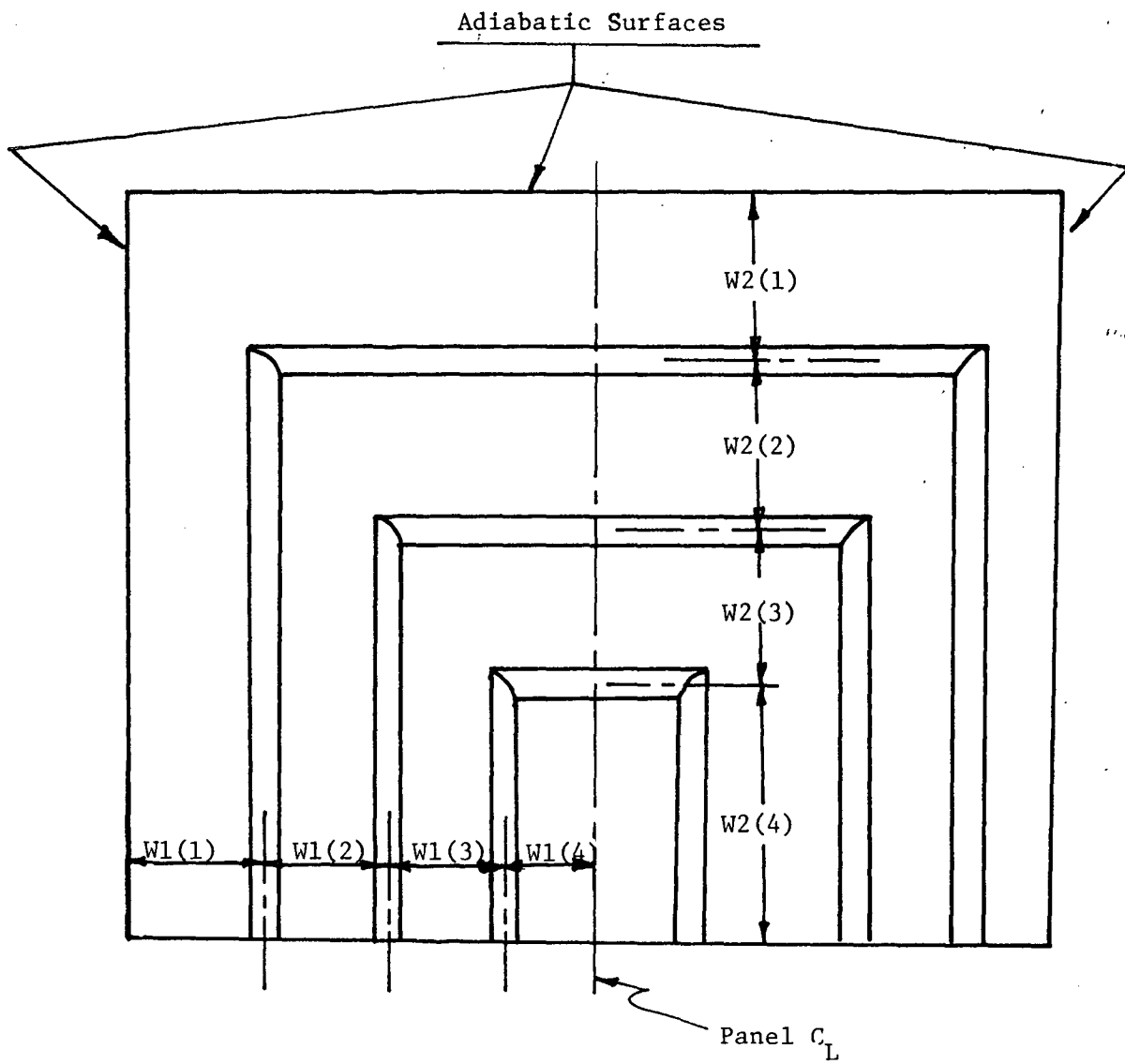


Fig. 4 The Dimensions  $W1(I)$  and  $W2(I)$  for U Shaped Coolant Channels

NTM            number of instances at which incident flux values  
 on to be specified if either card or file data is  
 used;  $NTM \leq 100$ . If ITAPE > 0, NTM must be the  
 number of data values stored on the MRI file

TØ            time in minutes during orbit at which irradiation  
 data is required. Time is measured from zero time  
 specified on MRI output data. If ITAPE = 0 this  
 value is ignored

PHIN           array of NT values of the angles in degrees  
 between the normals to the NT flat radiator  
 segments and the reference line used as the zero  
 angle reference on the MRI program

c) Ascent and Reentry Data

Velocity and altitude profiles for the shuttle vehicle during both ascent and reentry are read from cards which have the format listed in the table below. Selection of ascent or reentry phases is determined by the control integer NCØNV (see Section (e) which explains the Execution Control Options).

Card No.	FORMAT	Variable
1	2I10	NA, NR the number of ordered pairs of elements in the velocity-time and altitude-time arrays for ascent and reentry profiles, respectively
$\begin{matrix} 2 \\ \text{through} \\ INT^* (NA/10 + 0.9) \\ + 1 = N_1 \end{matrix}$	10F8.3	TA, elements of time array (NA values) in seconds, selected for ascent velocity and elevation profiles
$\begin{matrix} N_1 + 1 \\ \text{through} \\ N_1 + INT(NR/10 + 0.9) \\ = N_2 \end{matrix}$	10F8.3	TR, elements of time array (NR values) in seconds, selected for reentry velocity and elevation profiles

\*  $INT(x)$  = Integer part of x

$N_2 + 1$ through $N_2 + \text{INT}(NA/8+0.875)$ = $N_3$	8E10.3	VELA, elements of velocity array (NA values) in ft/sec, selected for ascent velocity profile of orbiter
$N_3 + 1$ through $N_3 + \text{INT}(NR/8+0.875)$ = $N_4$	8E10.3	VELR, elements of velocity array (NR values) in ft/sec, selected for reentry velocity profile of orbiter
$N_4 + 1$ through $N_4 + \text{INT}(NA/8+0.875)$ = $N_5$	8E10.3	ALTA, elements of altitude array (NA values) in ft, selected for ascent elevation profile of orbiter
$N_5 + 1$ through $N_5 + \text{INT}(NR/8+0.875)$	8E10.3	ALTR, elements of altitude array (NR values) in ft, selected for reentry elevation profile of orbiter

#### d) Irradiation Data

Incident radiant flux on the space base during orbit may be either read from card images or it may be read from a user supplied data file (MRI program). The selection of card or file input is determined by the value of the control integer ITAPE. If ITAPE = 0, the irradiation data is read from cards in the format shown below. If ITAPE > 0, the irradiation data is read from unit ITAPE. The data is assumed to be in the format specified in the MRI manual "Orbiting Satellite Surface Temperature Prediction and Analysis," MRI Project No. NAS9-1059.

The input variables used to input the irradiation data are to be supplied in the NAMELIST format called QNML, if ISYM = 0. The input strings must be in the sequence shown below. If ISYM ≠ 0, these values are supplied from the NAMELIST called MRI

NAMELIST/QNML/contains 5 variables

ITAPE = 0 to read irradiation data from cards

> 0 to read irradiation data from MRI tape. Value of ITAPE designates the input unit used to read file data

ICASE integer used to designate the case number of stored data to be used for input. ICASE = 3,

for example, would designate the third set of data stored on the MRI file as the irradiation data to be used for current input. If ITAPE = 0, this integer is ignored

- NTM            number of instances at which incident fluxes are to be specified if either card or file data is used;  $NTM \leq 100$ . If ITAPE > 0, NTM must be the number of data values stored on the MRI file
- TØ            time in minutes during orbit at which irradiation data is required. Time is measured from zero time specified on MRI output data. If ITAPE = 0 this value is ignored
- PHN            single angle in degrees between normal to panel and reference line used as a zero angle reference in MRI program.

If the irradiation data is to be read from cards, it must be in the format shown below with the cards immediately following the NAMELIST input cards for QNML.

Card No.	FORMAT	Variable											
1 through NTM	10F8.3	<p>On each card:</p> <table style="border: none;"> <tr> <td style="padding-right: 10px;">(i) time TM in seconds</td> <td rowspan="4" style="font-size: 3em; padding: 0 10px;">}</td> <td rowspan="4" style="vertical-align: middle;">irradiation (Btu/hr ft<sup>2</sup>)</td> </tr> <tr> <td>(ii) 3 values of solar</td> </tr> <tr> <td>(iii) 3 values of albedo</td> </tr> <tr> <td>(iv) 3 values of planetary</td> </tr> </table> <p>Incident at TM and each coming from one of three directions in this order:</p> <table style="border: none;"> <tr> <td style="padding-right: 10px;">(i) -30°</td> <td rowspan="3" style="font-size: 3em; padding: 0 10px;">}</td> <td rowspan="3" style="vertical-align: middle;">from the panel normal (upper side)</td> </tr> <tr> <td>(ii) 0°</td> </tr> <tr> <td>(iii) +30°</td> </tr> </table>	(i) time TM in seconds	}	irradiation (Btu/hr ft <sup>2</sup> )	(ii) 3 values of solar	(iii) 3 values of albedo	(iv) 3 values of planetary	(i) -30°	}	from the panel normal (upper side)	(ii) 0°	(iii) +30°
(i) time TM in seconds	}	irradiation (Btu/hr ft <sup>2</sup> )											
(ii) 3 values of solar													
(iii) 3 values of albedo													
(iv) 3 values of planetary													
(i) -30°	}	from the panel normal (upper side)											
(ii) 0°													
(iii) +30°													
NTM + 1 through 2*NTM	10F8.3	Same data as above, now for lower fin panel side, only if NSRAD = 2											

e) Coolant Fluid Temperature and Mass Flux Histories

When the temperature and mass flow rates entering the coolant channels vary with respect to time, the variation in these quantities are entered from cards in the format indicated below.

Card No.	FORMAT	Variable
1	I10	NFLDTA the number of data values of temperature and mass flow rate to be entered on each card:
2 through NFLD + 1	3F20.6	i) TMEFLD the time in hours corresponding to the given temperature and mass flow rate ii) TIFLD coolant fluid temperature in R at time NFLDTA iii) WIFLD coolant mass flow rate in lbm/hr at time NFLDTA

f) System Specifications

The system specifications define the radiator system and follow immediately after the ascent and coolant fluid temperature and mass flux histories.

The first four cards in the group of the system specification data contain the names of

- (i) the tube material
- (ii) the fin material
- (iii) the coolant fluid
- (iv) the meteoroid protection layer material

in alphanumeric form, beginning with an alphabetic character. Each name is punched on a separate card, beginning in the first column and extending no further than through the first twelve columns. These names serve to identify the printout of results. The user must supply these names in agreement with the composition of the Selective Part described in Chapter I.A. The subsequent data records contain numerical data.

(i) NAMelist/TUBE/contains 7 variables

DITBI internal (hydraulic) tube diameter, in inch,  
 STBI tube wall thickness, in inch,  
 XL tube length, in ft,  
 RHØTBI tube wall density, in lbm/ft<sup>3</sup>,  
 MZ number of nodal points along the tube axis  
 (no. of intervals plus one), ≤ 10,  
 NRTBI number of nodal points in radial direction, in  
 the tube wall, ≤ 5,  
 NTBS number of flow channels.

(ii) NAMelist/FLOW/contains 3 variables

MDØTI total coolant mass flow rate, in lbm/hr,  
 TO entrance (and reference) temperature in degrees  
 R,  
 PO entrance (and reference) pressure in lbf/ft<sup>2</sup>.

(iii) NAMelist/FIN/contains 7 variables

NX number of nodal points on the fin, perpendicular  
 to the channel axis, NX ≤ 10,  
 SRØØTI fin root thickness, in inch,  
 HFNI fin height, from root to tip, in inch,  
 STIPI fin tip thickness, in inch,  
 RHØFNI fin material density, in lbm/ft<sup>3</sup>,  
 STAGX distance from stagnation point on shuttle vehicle  
 to the center of radiator system measured along  
 a streamline, in ft,  
 VERTX total dimension of radiator panel in the direction  
 parallel to the acceleration of gravity, in ft.

(iv) NAMelist/PRØTLR/contains 13 variables

NRMP number of radial nodal points in the protection  
 layer, ≤ 5,  
 RHØMPI protection layer material density, in lbm/ft<sup>3</sup>,  
 RHØMET meteoroid density in g/cm<sup>3</sup>,  
 VELM meteoroid velocity in ft/sec,  
 TAU time that vulnerable area is exposed to meteoroid  
 environment, in days,  
 PRØB probability of no damage caused by meteoroid  
 impacts, dimensionless,



ALPHA	experimental constant that relates meteoroid flux and mass (see Eq. 8.4) in $\text{gm}/(\text{day ft}^2)$ ,
BTA	experimental constant that relates meteoroid flux and mass (see Eq. 8.4) dimensionless,
GAMMA	empirical constant used to adjust predicted penetration depths to one observed experimentally (see Eq. 8.1) dimensionless,
PHI	empirical constant (see Eq. 8.1) dimensionless,
THETA	empirical constant (see Eq. 8.1) dimensionless,
ATK	empirical constant used to account for spalling on a target of finite thickness (see Eq. 8.8) dimensionless,
AN	experimental constant that describes penetration depth as a function of angle of incidence (see Eq. 8.6) dimensionless.

(v) NAMELIST/MANIFD/contains

AMAN	total manifold area, projected into the fin plane, in $\text{ft}^2$ .
------	---

Any one of these variables may be changed for successive program executions by entering them in NAMELIST/GINPT/ as discussed below.

g) Execution Control Options

The user has the choice of options for the execution of the first as well as successive computer runs.

Two data records are considered to be execution controls: RUNOPT and GINPT.

(vi) NAMELIST/RUNOPT/contains 11 variables

MSTOPT	= 1 to compute steady state conditions
	= 2 to simulate transient system performance,
DTWRTE	fixed time interval, in hr, between data printout during integration,
TEND	termination time, in hr, for transient performance calculation,
ALIMIT	absolute error limit per integration step, see Eq. 15.3, in Systems Analysis Manual

RLIMIT      relative error limit per integration step, see Eq. 15.3, in Systems Analysis Manual.

TI            initial temperature, in degrees R,

LIMWRT      maximum number of data recording during integration toward steady state, exclusive of initial conditions record and steady state record,

NCØNV      = 0    no aerodynamic heating  
               = 1    ascent  
               = 2    reentry,

LTT          = 0    optical properties of the surface coating are independent of temperature  
               = 1    optical properties of the surface coating are dependent on temperature  
               = 2    irradiation on fin on tube is zero but subroutine QINCID is called,

LFLD        = 1    coolant fluid inlet temperature and mass flow rate are constant  
               = 2    temperature and mass flow rate of inlet coolant fluid is variable,

LTS          = 0    uniform initial system temperature equal to TI  
               = 1    initial system temperature taken to be equal to previously obtained values.

(vii) NAMelist/GINPT/contains every variable name listed in items (i) through (vi) in addition to NSRD and the following control integer.

MRSTR1 = 1    new velocity and altitude profiles  
           = 2    new irradiation history  
           = 3    both new velocity, altitude profiles plus new irradiation history  
           = 4    new coolant fluid inlet conditions  
           = 5    new coolant fluid inlet conditions, plus new velocity and altitude profiles  
           = 6    new coolant fluid inlet conditions, plus new irradiation history  
           = 7    new coolant fluid inlet conditions, plus new velocity and altitude profiles, plus new irradiation history  
           = 8    the only new input variables are those specified in GINPT.

There may be as many NAMELIST/GINPT/ records as computing time permits. If GINPT is omitted, the program will terminate.

## 4. Results

### a. Information Produced

The results produced by the program can be divided into three categories each of which is listed on a separate output page. All input variables that specify the system and select the program options are printed on the first page of output (see Fig. 5 ). The second page contains all initial line conditions and system parameters which remain constant during the integration process (see Fig. 6 ). The succeeding pages contain the current system description printed during the integration process (see Fig.7 ). If the system is non-symmetrical an additional page of output is printed. This page (see Fig. 8 ) lists system parameters when the radiator panel is curved, the flow channels are not identically loaded, or the flow channels are U-shaped.

### b. Output Format

The first output record consists of the list of all system specifications and execution control data as listed in Sections 3(d) and 3(e). This output appears with all of the variables in the NAMELISTS TUBE, FLOW, FIN, PRØTLR, MANIFD and RUNØPT listed in the NAMELIST format. A sample of this output is shown in Fig. 5 .

On the second page of output the initial coolant fluid flow properties are printed in non-dimensional form (see Fig. 6 ). This information includes non-dimensional pressure, velocity, temperature, and tube wall temperature. The reference quantities are listed below the initial line conditions. They are inlet fluid pressure, velocity and temperature. Next are printed several dimensionless parameters such as the Reynold's number, Prandtl number and Nusselt number for the initial coolant fluid flow conditions. The quantity DELTA is the ratio of the internal tube diameter to the tube length. The relative pressure is given by equation 3.31 in the System Analysis Manual. The Biot number of the tube is also printed.

0XGT N.VER1	
\$TUSE	
DITBI =	.2500000E+00
STBI =	.1000000E+00
XL =	.2000000E+01
RHOTBI =	.1685200E+03
MZ =	+5
NRIBI =	+3
NTBS =	+12
SEND	
\$ELOW	
MDOYI =	.50001000E+03
ID =	.7000000E+03
PO =	.25001000E+04
SEND	
\$EIN	
NX =	+5
SROOYL =	.5000000E-01
HFNI =	.1200000E+02
STIPI =	.5000000E-01
RHOJNI =	.1685200E+03
SIAGX =	.2000000E+02
VERTX =	.1000000E+02
SEND	
\$PROTLR	
NRMP =	+3
RHOMPI =	.11357400E+03
RHOMET =	.5000000E+00
VELM =	.6560000E+05
TAU =	.3650000E+04
PROB =	.9900000E+00
ALPHA =	.1880000E-09
BTA =	.1213000E+01
GAMMA =	.1500000E+01
PHI =	.5000000E+00
THETA =	.66666700E+00
ATK =	.1750000E+01
AN =	.1000000E+01
SEND	
\$MANIFD	
AMAN =	.4300000E+02
SEND	
\$RUNOPT	
MSTOTR =	+2
DIRTE =	.8333000E-02
TEND =	.2000000E+00
ALIMIT =	.5000000E-04
RLIMIT =	.1000000E-04
TI =	.6600000E+03
LIMVRT =	+20
NCONV =	+0
LIT =	+0

FIGURE 5

INPUT VARIABLES

LFLD =  
LTS =  
\$END

+2  
+0

02

INITIAL LINE CONDITIONS  
 \*\*\*\*\*  
 (ALL QUANTITIES ARE NORMALIZED)

PT. NO.	POSITION Z	PRESSURE P	VELOCITY W	FLUID TEMPERATURE T	WALL TEMPERATURE TWI
1	.000	1.000000	.974335	.942857	.942857
2	.250	.998340	.974339	.942857	.942857
3	.500	.996680	.974342	.942858	.942857
4	.750	.995020	.974346	.942858	.942857
5	1.000	.993359	.974350	.942858	.942857

INLET PRESSURE P0 = 2500.000 LBF/SQ.FT  
 REF. VELOCITY W0 = .69331 FT/SEC  
 REF. TEMPERATURE T00 = 700.000 R

REYNOLDS NO = .13191+03  
 PRANDTL NO = 95.079041  
 DELTA = .010417  
 REL. PRESSURE IS = .170849+04

INIT. NUSSLETT NO. NU = .768940+01  
 WALL BIOT NO. BI = .362036-02

SYSTEM PARAMETERS  
 \*\*\*\*\*

TUBE LENGTH , XL = 2.000 FT  
 INTERNAL DIAMETER , DIIB = .250 IN  
 WALL THICKNESS , STB = .100 IN  
 MATERIAL = ALUMINUM  
 MASS (ALL TUBES) , MTB = 3.0896 LBM  
 NUMBER OF TUBES , NTBS = 12

FIN HEIGHT , HFN = 12.000 IN  
 THICKNESS AT ROOT , SROOT = .050 IN  
 THICKNESS AT TIP , STIP = .050 IN  
 MATERIAL = ALUMINUM  
 MASS (ALL FINES) , MFN = 33.7184 LBM  
 NO. OF FIN SIDES RADIATING = 1

COOLANT FLUID IS SILICONE OIL  
 MASS (IN ALL TUBES) , MFL = 4.9345 LBM

PROTECTION LAYER THICKNESS, SMP = .063 IN  
 MASS, MMP = 1.804 LBM  
 MATERIAL IS BERYLLIUM

TOTAL MASS (EXCL. MANIFLD.) MTOT = 43.5462 LBM  
 TOTAL AREA (SINGLE NORMAL PROJECTION) , ATOT = 48.9000 SQ FT

FIGURE 6  
 INITIAL LINE CONDITIONS AND SYSTEM PARAMETERS

ELAPSED TIME IS .0000 HR  
 RELATIVE TIME IS .0000  
 \*\*IN ORBIT\*\*  
 I INTEGR. STEPS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*  
 REFERENCE TEMPERATURE, T00 = 700.000 R  
 REF. RADIANT HEAT FLUX, QREF = .9877+04 BTU/(HR.FT), INCIDENT SOLAR FLUX, QSOLR = .1000+03 BTU/(HR SQ FT)  
 PER UNIT AXIAL LENGTH, GREF = .1295+05 BTU/HR, INCIDENT INFRARED FLUX, QIR = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION, GTOT = .0000 BTU/HR, AERODYN. HEATING POWER, GCONV = .0000 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .0000 BTU/HR, ENERGY STORAGE RATE, STORG = -.1295+05 BTU/HR

AXIAL DIST.	RELATIVE RAD. HEAT REJECTION	RELATIVE TEMPERATURE OF FIN, T	DISTANCE NORMAL TO FLOW DIRECTION X
0.0000	.25000	.50000	.75000 1.00000
.000	.6564	.9429	.9429
.250	.6552	.9429	.9429
.500	.6551	.9429	.9429
.750	.6552	.9429	.9429
1.000 (EXIT)	.6564	.9429	.9429

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*  
 REFERENCE TEMPERATURE, T00 = 700.000 R  
 REFERENCE PRESSURE, P00 = 2500.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .693 FT/SEC  
 COOLANT POWER, INLET AT I=ID H0 = .433766+05 BTU/HR  
 EXIT CURRENTLY E1 = .361473+05 BTU/HR  
 INLET CURRENTLY HI = .361472+05 BTU/HR  
 TOT. REJECTION DH = -.108887+00 BTU/HR

AXIAL DIST.	PROTECT. LAYER TEMPERATURE T	WALL TEMPERATURE TW	FLUID TEMPERATURE TF	VELOCITY W	ENTHALPY REJECTION PER UNIT TUBE LENGTH BTU/(HR.FT)	FRACTION OF TOTAL
.000	.9429	.9429	.9429	.9429	.000057	.00000
.250	.9429	.9429	.9429	.9429	.000283	.05725
.500	.9429	.9429	.9429	.9429	.000548	.25331
.750	.9429	.9429	.9429	.9429	.000803	.56830
1.000 (EXIT)	.9429	.9429	.9429	.9429	.001039	1.00000

FIGURE 7  
 CURRENT SYSTEM DESCRIPTION



ELAPSED TIME IS .0083 HR  
 RELATIVE TIME IS 10.3992  
 \*\*IN ORBIT\*\* 23 INTEGR. STEPS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION

REFERENCE TEMPERATURE, T00 = 700.000 R  
 REF. RADIANT HEAT FLUX, QREF = .9877+04 BTU/(HR.FT), INCIDENT SOLAR FLUX, QSOLR = .1000+03 BTU/(HR SQ FT)  
 PER UNIT AXIAL LENGTH, GREF = .1198+05 BTU/HR, INCIDENT INFRARED FLUX, QIRED = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION, GTOT = .4684+03 BTU/HR, AERODYN. HEATING POWER, QCONV = .0000 BTU/HR  
 COND. FROX MANIFOLDS, CONDFE = .4684+03 BTU/HR, ENERGY STORAGE RATE, STORG = -.1085+05 BTU/HR

AXIAL RELATIVE RAD. HEAT REJECTION

DIST. Z DISTANCE NORMAL TO FLOW DIRECTION

Z	0	.0000	.25000	.50000	.75000	1.00000
.000	.6304	.9342	.9342	.9342	.9342	.9342
.250	.6024	.9332	.9248	.9233	.9232	.9233
.500	.6010	.9331	.9244	.9228	.9226	.9228
.750	.6022	.9331	.9248	.9233	.9232	.9233
1.000 (EXIT)	.6292	.9338	.9338	.9338	.9338	.9338

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER

REFERENCE TEMPERATURE, T00 = 700.000 R  
 REFERENCE PRESSURE, P00 = 2500.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .693 FT/SEC  
 COOLANT POWER, INLET AT T=10, H0 = .433766+05 BTU/HR  
 EXIT CURRENTLY EI = .378522+05 BTU/HR  
 INLET CURRENTLY HI = .385224+05 BTU/HR  
 TOT. REJECTION DH = .670021+03 BTU/HR

AXIAL DIST.	PRESSURE P	VELOCITY W	TEMPERATURES			PER UNIT TUBE LENGTH BTU/(HR.FT)	ENTHALPY REJECTION FRACTION OF TOTAL
			FLUID TF	WALL TWI	LAYER TMP		
.000	1.00000	1.0046	.9539	.9342	.9342	25.000915	.00000
.250	.99824	1.0046	.9525	.9332	.9332	24.412890	.27294
.500	.99648	1.0046	.9512	.9331	.9331	22.919608	.53364
.750	.99473	1.0046	.9499	.9331	.9331	21.360966	.77689
1.000 (EXIT)	.99297	1.0046	.9488	.9338	.9338	18.983883	1.00000

FIGURE 7 CONCLUDED

FLAT PANEL  
 3 STRAIGHT TUBES  
 2 RADIATING SIDES

TUBE SPECIFICATIONS 2.5000-01 INCHES INSIDE DIAMETER

TUBE LENGTH (FEET)  
 2.0000+00 2.0000+00

FIN SPECIFICATIONS

FIN THICKNESS (INCHES)  
 5.0000-02 5.0000-02 5.0000-02

FIN HALF-WIDTH (INCHES)  
 1.1000+01 1.1500+01 1.2000+01 1.2500+01

SINK TEMPERATURE (DEGREES R)  
 4.9391+02 4.9391+02 4.9391+02  
 5.0736+02 5.0736+02 5.0736+02

FLUID SPECIFICATIONS

INLET TEMPERATURE (DEGREES P)  
 7.0000+02 7.0000+02 7.0000+02

INLET PRESSURE (LBF/SG.FT.)  
 2.5000+03 2.5000+03 2.5000+03

MASS FLOW RATE (LBM/HR)  
 5.0000+02 5.0000+02 5.0000+02

FIGURE 8  
 SYSTEM PARAMETERS FOR NON-SYMMETRICAL HEATING

TUBE	BULK TEMP DEG R		INCHES TO ADIABATIC PLANE		PER CENT FIN EFFICIENCY		FIN TIP TEMP		OUTLET TEMP		HEAT REJECTED BTU/HR
	1	2	LEFT	RIGHT	LEFT	RIGHT	LEFT	RIGHT	LEG R	DEG R	
1	6.0619+02	6.0970+02	1.1000+01	1.1531+01	5.9360+01	5.7589+01	5.7493+02	5.7314+02	6.0619+02	6.0619+02	1.4465+03
2	6.0970+02	6.0524+02	1.1462+01	1.2028+01	5.7816+01	5.6025+01	5.7314+02	5.7135+02	6.0570+02	6.0570+02	1.4542+03
3	6.0524+02	6.0970+02	1.1972+01	1.2500+01	5.6226+01	5.4593+01	5.7135+02	5.6975+02	6.0524+02	6.0524+02	1.4613+03

A listing of system parameters follows the initial line condition. The system parameters include the pertinent geometry parameters, the materials and the mass of the tube, fin, coolant fluid and protection layer materials. In addition, the thickness of the meteoroid protection layer, the total system mass excluding the mass of the manifolds and the total projected areas are listed at the bottom of the page.

Following the preliminary results is a series of current system descriptions printed out during the integration process (see Fig.7 ). The first items printed are the current real time and the relative time which is the ratio of the current time to the time required for a fluid particle to pass through the flow channel. If the calculations are for the space base in orbit, the words "in orbit" are printed at the top of the page. If the calculations are for either ascent or reentry of the space shuttle, the current shuttle velocity and altitude are printed along with the current atmospheric temperature. For either the sub-orbital or orbital case, the cumulative number of integration steps from the initial conditions is printed at the top of the page.

The next data printed are a summary of the heat flux terms and a listing of the non-dimensional fin temperature distribution. The reference temperature is the inlet fluid temperature and the reference radiant heat flux is the black body emissive power at the inlet coolant fluid temperature evaluated for a unit length of fin and tube segment. The total radiant rejection and the conduction into the fin from both the inlet and exit manifolds for the entire radiator system are also printed. The external heat fluxes incident on the system that are printed are the infrared irradiation from earth albedo, solar irradiation and the convective flux on the radiator system of the shuttle vehicle during ascent and reentry. The energy storage rate for the entire system is also listed along with the external heat fluxes.

The final table listed on the page contains coolant fluid properties as functions of axial distance as well as the local cooling rates. All properties are tabulated in non-dimensional form and the table is preceded by a list of the necessary reference quantities. Properties of the coolant fluid that are printed are pressure, velocity and bulk temperature. The

non-dimensional interior tube wall temperature and the normalized temperature of the meteoroid protection layer temperature are also listed as functions of axial distance down the tube. The enthalpy rejection of the fluid and fraction of the total enthalpy drop are listed as functions down the tube. The latter parameter is helpful in determining the effectiveness of the various segments of the tube length in their capacity to reject heat. Below the reference properties used for the fluid properties are the inlet and exit fluid stagnation enthalpies as well as the total change in stagnation enthalpy. These quantities are labeled respectively, HI, HE and DH. Also for comparison is listed the enthalpy flux at the reference state ( $T_0, P_0$ ) and the reference velocity  $W_0$ .

The last table of current output data constitutes either the system conditions at the specified termination time, TEND in hours, or the steady state condition depending on whether MSTØTR = 1 or MSTØTR = 2, respectively. For conditions of steady state, the words "STEADY STATE IS REACHED" are printed after the final data printout.

When input parameters lead to a condition for which the tube centerline or the midplane separating two tubes are not lines of symmetry, then a single page of output is printed prior to all other output. This page contains a written description of tube arrangement on the radiator and a printout describing the panel as either flat or curved. The number of non-symmetrical tubes and the number of sides of the panel radiating are also printed at the top of the page.

The next output consists of a description of the sink temperature data including orientation of the fin segments relative to the reference vector used in the MRI program.

The tube specifications follow the sink temperature data. The data listed are the internal tube diameter and the tube lengths when the tubes are straight. If the tubes are U-shaped, the spacings at the inlet and outlet segments as well as the crossover segments of the tubes are listed instead of the tube lengths.

The fin specifications follow the tube data. The data listed are fin thicknesses and the distance between the tube centerline and the midplane separating two tubes. The effective sink temperature for each fin segment is also printed under fin specifications. When both sides of the panel radiate, two lines of sink temperatures are printed corresponding to the sink temperature both radiator sides.

The coolant fluid specifications follow the fin data. Inlet fluid temperature, pressure and mass flow rates are printed for each non-symmetrical tube.

The final data to be printed on the page is a summary of the system conditions for all of the non-symmetrical fin segments. The data printed includes the fin bulk temperature, position of the adiabatic plane, the fin effectiveness, fin tip temperature, fluid outlet temperature and total heat rejected by the fluid in each of the tubes.

This single page of output is then followed by the usual system printout as described above, repeated once for each non-symmetrical tube.

### c. Output Listing

Figures 5, 6 and 7 in this section are typical output listings produced by the simulation program. Figure 5 provides a listing of all input parameters that appear in the NAMELIST format. Figure 6 shows initial line conditions and system parameters and Figure 7 shows a typical printout of the current radiator system parameters. These three figures are typical of program output when the radiator panel is symmetrical about each flow channel. If non-symmetrical conditions exist, output similar to that shown in Figure 8 is produced.

## 5. Typical Radiator System Performance

This section summarizes typical program results for expected radiator input conditions during ascent and reentry for the shuttle vehicle and during both transient and steady state performance of the radiator system in a simulated orbit.

### a. Ascent

#### (i) Mission Description

The ascent altitude and velocity profiles used for the system output shown in Figures 10 and 11 is given in Figure 9. Other input parameters were selected as reasonable conditions expected for a typical space shuttle radiator system. The total fin area of 124.5 ft<sup>2</sup> is subdivided by 10 equally spaced 12-foot long flow channels. The total fin height between flow channels is 1 ft. The inlet coolant fluid temperature and pressure are 200 F and 60 lb<sub>f</sub>/in<sup>2</sup>, respectively. The system materials are aluminum tubes, fins and meteoroid protection material with silicon oil as the coolant fluid. Other system parameters are printed as input listing which precedes the current system parameters shown in Figure 10a. The panel radiates only from one side. Incident radiant fluxes are constant during ascent at 100 Btu/hr ft<sup>2</sup> for solar irradiation, 2.0 Btu/hr ft<sup>2</sup> for earth albedo. There is no infrared earth irradiation.

#### (ii) System Response

The response of the radiator system during ascent is summarized in Figures 10 and 11. A selected number of output listings of current system parameters during ascent are shown in Figure 10a. Figure 10b is a plot of the total heat rejection from the entire radiator surface during ascent as well as the outlet coolant fluid temperature. Figure 11 is a plot of the aerodynamic heating at various times during the ascent phase of the mission.

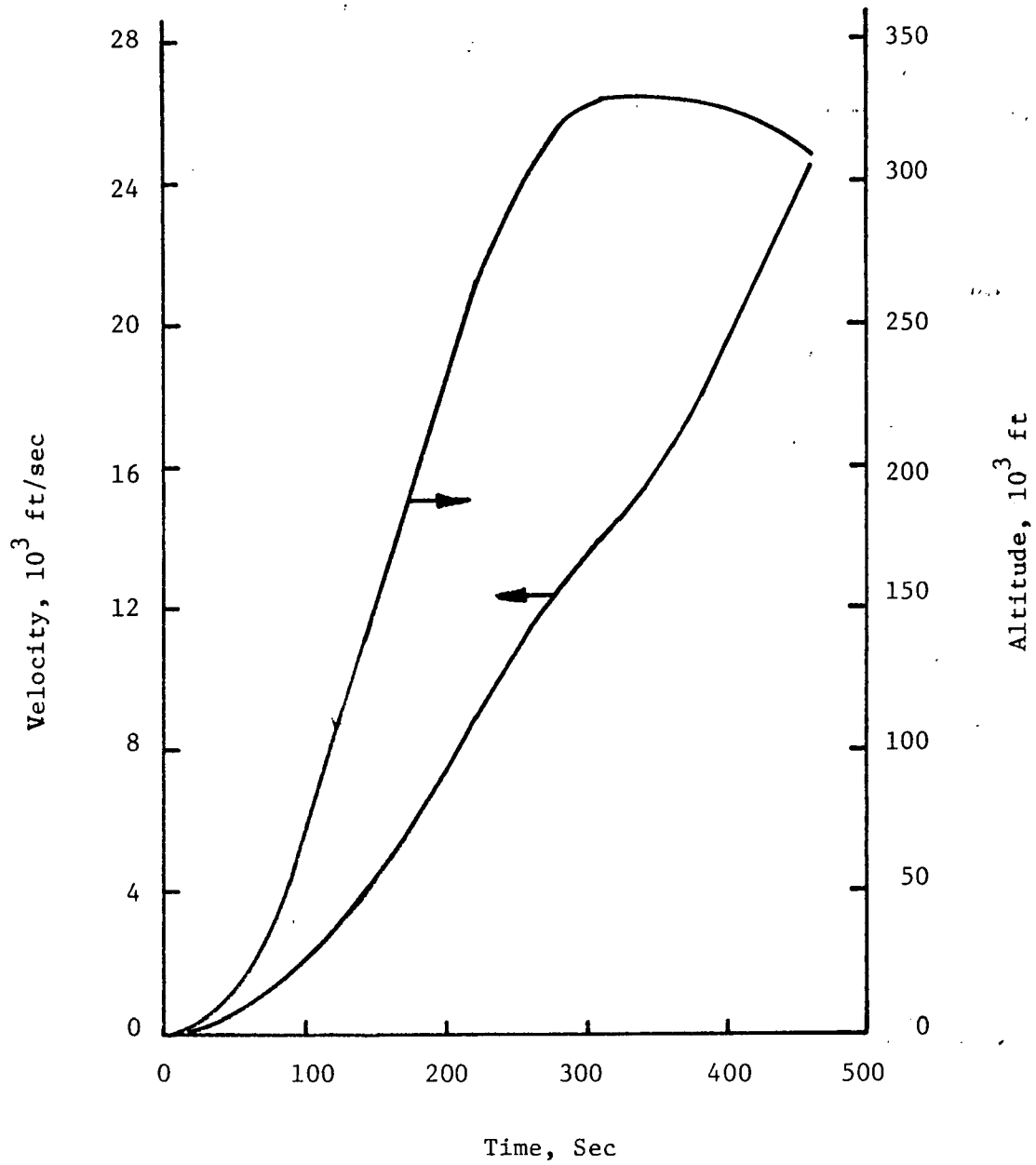


Fig. 9 Ascent Velocity and Altitude Profiles

```

$QNML = +0
ITAPE = +0
ICASE = +0
NIN = +4
TO = +0000000E+00
PHN = +00000000E+00

SEND
$TUBE = +2500000E+00
DITBI = +1000000E+00
STBI = +1200000E+02
XL = +16859200E+03
RHOtBI = +9
Nz = +3
NRTBI = +10
ITBS = +10

SEND
$FLOW = +5000000E+03
TO = +6596700E+03
PO = +8640000E+04

SEND
$FIN = +5
NX = +5000000E-01
SRO01 = +6000000E+01
HFNI = +5000000E-01
STPI = +16859200E+03
RHOtBI = +2000000E+02
STAGA = +1000000E+02
VERTX = +1000000E+02

SEND
$PROTLR = +3
NRMP = +16859200E+03
RHOMP1 = +5000000E+00
RHOMET = +6600000E+05
VELM = +36500000E+04
TAU = +99000000E+00
PKOB = +1880000E-09
ALPHA = +12130000E+01
BTA = +1500000E+01
GAMMA = +5000000E+00
PHI = +6666700E+00
THETA = +1750000E+01
ATK = +1000000E+01
AN = +1000000E+01

SEND
$MANIFD = +3000000E+02
AMAN = +3000000E+02

SEND
$RUNUPT = +2
MSTOIR = +2

```

Figure 10(A) SYSTEM RESPONSE FOR ASCENT  
PROGRAM OUTPUT



DTWRT = .8330000E-02  
 TEND = .1270000E+00  
 ALIMIT = .5000000E-04  
 RLIMIT = .1000000E-04  
 Y1 = .6596700E+03  
 LIMWT = +25  
 NCONV = +1  
 LTT = +0  
 LFLO = +1  
 LTS = +0  
 SEND

FIGURE 10(A) CONTINUED

16211

INITIAL LINE CONDITIONS

\*\*\*\*\*

ALL QUANTITIES ARE NORMALIZED

PT. NO.	POSITION Z	PRESSURE P	VELOCITY W	FLUID TEMPERATURE T	WALL TEMPERATURE TWI
1	.000	1.000000	1.000000	1.000000	1.000000
2	.125	.999810	1.000000	1.000000	1.000000
3	.250	.999621	1.000001	1.000000	1.000000
4	.375	.999431	1.000001	1.000000	1.000000
5	.500	.999241	1.000001	1.000000	1.000000
6	.625	.999052	1.000001	1.000000	1.000000
7	.750	.998862	1.000001	1.000000	1.000000
8	.875	.998673	1.000002	1.000000	1.000000
9	1.000	.998483	1.000002	1.000000	1.000000

INLET PRESSURE P0 = 8640.000 LBF/50.FT  
 REF. VELOCITY W0 = .87128 FT/SFC  
 REF. TEMPERATURE T00 = 659.670 R

REYNOLDS NO = .30017+04  
 PRANDTL NO = 8.628066  
 DELTA = .001736  
 REL. PRESSURE IS = .391534+04

INIT. NUSSFLT NO. NU = .201856+02  
 WALL BIOT NO. BI = .485314-02

FIGURE 10(A) CONTINUED

SYSTEM PARAMETERS

\*\*\*\*\*

TUBE LENGTH, XL = 12.000 FT  
 INTERNAL DIAMETER, DITB = .250 IN  
 WALL THICKNESS, STB = .100 IN  
 MATERIAL = ALUMINUM  
 MASS (ALL TUBES), MTD = 15.4480 LBM  
 NUMBER OF TUBES, NTBS = 10  
 FIN HEIGHT, HFN = 6.000 IN  
 THICKNESS AT ROOT, SROOT = .050 IN  
 THICKNESS AT TIP, STIP = .050 IN  
 MATERIAL = ALUMINUM  
 MASS (ALL FIN), MFN = 84.2960 LBM  
 NO. OF FIN SIDES RADIATING = 1

COOLANT FLUID IS SILICONE OIL  
 MASS (IN ALL TUBES), MFL = 19.1289 LBM

PROTECTION LAYER THICKNESS, SMP = .011 IN  
 MASS, MMP = 2.038 LBM  
 MATERIAL IS BERYLLIUM

TOTAL MASS (EXCL. MANIFLO.) MTOT = 120.9104 LBM

TOTAL AREA (SINGLE NORMAL PROJECTION) ,  
ATOT = 124,5000 SQ FT

62

FIGURE 10(A) CONTINUED

ELAPSED TIME IS .0000 HR , RELATIVE TIME IS .0000 1 INTEGR. STEPS  
 ALTITUDE IS .00 FT , VELOCITY IS .00 FT/SEC , ATM. TEMPERATURE IS 518.67 R

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, GSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT) , INCIDENT INFRARED FLUX, QIRED = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION QTOT = .3289+05 BTU/HR , AERODYN. HEATING POWER, QCONV = -.1924+05 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .0000 BTU/HR , ENERGY STORAGE RATE, STORG = -.5213+05 BTU/HR

AXIAL RELATIVE DIST. RAD. HEAT REJECTION RELATIVE TEMPERATURE OF FIN, T

Z	Q	T	X	Y
.0000	.0000	.25000	.50000	.75000 1.00000
.000	.8477	1.0000	1.0000	1.0000 1.0000
.125	.8451	1.0000	1.0000	1.0000 1.0000
.250	.8436	1.0000	1.0000	1.0000 1.0000
.375	.8436	1.0000	1.0000	1.0000 1.0000
.500	.8436	1.0000	1.0000	1.0000 1.0000
.625	.8436	1.0000	1.0000	1.0000 1.0000
.750	.8436	1.0000	1.0000	1.0000 1.0000
.875	.8451	1.0000	1.0000	1.0000 1.0000
1.000	.8477	1.0000	1.0000	1.0000 1.0000

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .276866+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = .500488-01 BTU/HR

AXIAL DIST.	PRESSURE P	VELOCITY W	FLUID TEMP TF	WALL TEMP TMI	PROTECT. LAYER TEMP TPL	ENTHALPY REJECTION PER UNIT TUBE LENGTH BTU/(HR FT)	FRACTION OF TOTAL
.000	1.00000	1.0000	1.0000	1.0000	1.0000	.000000	.00000
.125	.99981	1.0000	1.0000	1.0000	1.0000	.000138	.02563
.250	.99962	1.0000	1.0000	1.0000	1.0000	.000242	.09431
.375	.99943	1.0000	1.0000	1.0000	1.0000	.000311	.19375
.500	.99924	1.0000	1.0000	1.0000	1.0000	.000380	.31625
.625	.99905	1.0000	1.0000	1.0000	1.0000	.000450	.46386
.750	.99886	1.0000	1.0000	1.0000	1.0000	.000484	.63147
.875	.99867	1.0000	1.0000	1.0000	1.0000	.000519	.80933
1.000	.99848	1.0000	1.0000	1.0000	1.0000	.000553	1.00000

ELAPSED TIME IS .0250 HR , RELATIVE TIME IS 6.5344 38 INTEGR. STEPS  
 ALTITUDE IS 56495.86 FT , VELOCITY IS 1449.87 FT/SEC , ATM. TEMPERATURE IS 389.97 R

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, GSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, GREF = .3246+04 BTU/(HR FT), INCIDENT INFRARED FLUX, GIRD = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION GTOT = .2115+05 BTU/HR , AERODYN. HEATING POWER, GCONV = -.1634+05 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .2443+04 BTU/HR , ENERGY STORAGE RATE, SIORG = -.3076+05 BTU/HR

AXIAL DIST. RELATIVE TEMPERATURE OF FIN, T  
 RAD. HEAT REJECTION DISTANCE NORMAL TO FLOW DIRECTION

Z	T	X	Y
.000	.0000	.25000	.50000
.125	1.2195	1.0879	1.0879
.250	.6981	.9517	.8047
.375	.3854	.8687	.8112
.500	.3856	.9572	.8110
.625	.5860	.8680	.8115
.750	.2817	.8691	.8115
.875	.3870	.9560	.8104
1.000	1.2263	.8697	.8117
(EXIT)		.8620	.8044
		1.0893	1.0893
		1.0893	1.0893

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .234041+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = .428260+04 BTU/HR

AXIAL DIST.	PROTECT. LAYER	WALL TEMPERATURE	FLUID TEMPERATURE	VELOCITY	PROTECT. LAYER TEMP	ENTHALPY REJECTION PER UNIT TUBE LENGTH	FRACTION OF TOTAL
		TF	TW	W	TEMP	BIU/(HR FT)	
.000	1.0000	1.0000	1.0879	1.0000	1.0879	-186.124290	.00000
.125	.9891	.9867	.9517	.9956	.9517	74.285834	-.11604
.250	.9262	.9871	.9572	.9928	.9572	63.325497	.50202
.375	.9942	.9760	.9550	.9913	.9550	44.420708	.52840
.500	.9923	.9815	.9574	.9896	.9574	50.998492	.75807
.625	.99704	.9689	.9560	.9891	.9560	27.315979	.98675
.750	.9885	.9881	.9585	.9881	.9585	46.887507	1.14491
.875	.9866	.9641	.9489	.9866	.9489	32.308490	1.38225
1.000	.99847	.9754	1.0893	.9970	1.0893	-241.334110	1.00000
(EXIT)							

FIGURE 10(A) CONTINUED

ELAPSED TIME IS .0500 HR , RELATIVE TIME IS 13.0687 55 INTEGR. STEPS 462.40 R  
 ALTITUDE IS 194986.66 FT , VELOCITY IS 6199.57 FT/SEC , ATM. TEMPERATURE IS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION

\*\*\*\*\*  
 REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, OSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT), INCIDENT INFRARED FLUX, QIRED = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION QTOT = .2358+05 BTU/HR , ALROJYN. HEATING POWER, QCONV = .2487+05 BTU/HR  
 CONJ. FROM MANIFOLDS, CONDMF = .2680+03 BTU/HR , ENERGY STORAGE RATE, STORG = .1333+05 BTU/HR

AXIAL DIST.	RELATIVE RAD. HEAT REJECTION	RELATIVE TEMPERATURE OF FIN, T	DISTANCE NORMAL TO FLOW DIRECTION X	RELATIVE TEMPERATURE OF METEOROID PROTECTION LAYER		
Z	0	0.0000	.25000	.50000	.75000	1.00000
.000	.7697	.9783	.9783	.9783	.9783	.9783
.125	.6604	.9172	.9160	.9152	.9147	.9147
.250	.5928	.9214	.9231	.9249	.9252	.9250
.375	.5873	.9133	.9177	.9216	.9231	.9233
.500	.5825	.9130	.9182	.9223	.9238	.9240
.625	.5725	.9076	.9160	.9225	.9254	.9261
.750	.5751	.9070	.9147	.9204	.9227	.9231
.875	.5904	.8927	.9012	.9077	.9106	.9112
1.000 (EXIT)	.6184	.9315	.9315	.9315	.9315	.9315

FIGURE 10(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER

\*\*\*\*\*  
 REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LRF/SQ.FI  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY E1 = .159106+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION OH = .117761+05 BTU/HR

AXIAL DIST.	PROTECT. LAYER	FLUID TEMPERATURE TF	WALL TEMPERATURE TW	PER UNIT TUBE LENGTH BTU/(HR FT)	ENTHALPY REJECTION FRACTION OF TOTAL
.000	1.0000	1.0000	.9783	42.947949	.00000
.125	.9981	.9927	.9814	126.966851	.14939
.250	.9956	.9874	.9750	106.037679	.33239
.375	.9942	.9827	.9603	92.991148	.47533
.500	.9923	.9784	.9561	85.266257	.61012
.625	.9904	.9748	.9429	69.837278	.72965
.750	.9885	.9714	.9070	68.399486	.83144
.875	.9866	.9677	.8927	71.936090	.93790
1.000 (EXIT)	.9847	.9672	.9315	-12.536525	1.00000

ELAPSED TIME IS .0750 HR , RELATIVE TIME IS 19.6031 80 INTEGR. STEPS 334.32 R  
 ALTITUDE IS 308492.96 FT , VELOCITY IS 11899.46 FT/SEC , ATM. TEMPERATURE IS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION

\*\*\*\*\*  
 REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, QSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX, PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT), INCIDENT INFRARED FLUX, QIRD = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION, QTOT = .2231+05 BTU/HR , AERODYN. HEATING POWER, QCONV = .6611+04 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .2343+03 BTU/HR , ENERGY STORAGE RATE, STORG = -.4995+04 BTU/HR

AXIAL RELATIVE RAD. HEAT REJECTION  
 DISTANCE NORMAL TO FLOW DIRECTION

RELATIVE TEMPERATURE OF FIN, T

0 .0000 .25000 .50000 .750\*0 1.00000

.000 .7313 .9671 .9671 .9671 .9671  
 .125 .6328 .9301 .9147 .9054 .9007 .8993  
 .250 .5703 .9322 .9199 .9128 .9093 .9084  
 .375 .5614 .9246 .9138 .9078 .9051 .9044  
 .500 .5533 .9224 .9127 .9075 .9051 .9046  
 .625 .5419 .9172 .9092 .9053 .9034 .9035  
 .750 .5395 .9146 .9068 .9030 .9015 .9012  
 .875 .5462 .9035 .8954 .8913 .8897 .8894  
 1.000 .5620 .9121 .9121 .9121 .9121 .9121  
 (EXIT)

FIGURE 10(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER

\*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LRF/59.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=10 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .172140+05 BTU/HR

INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = .1047721+05 BTU/HR

PROTECT. ENTHALPY REJECTION

FLUID TEMPERATURES PER UNIT TUBE LENGTH OF FRACTION

AXIAL DIST.	PRESSURE P	VELOCITY W	TF	WALL TEMPERATURES	PROTECT. LAYER TEMPS	PER UNIT TUBE LENGTH BTU/(HR FT)	FRACTION OF TOTAL
.000	1.00000	1.0000	1.0000	.9671	.9671	67.116564	.00000
.125	.99911	.9939	.9852	.9301	.9301	112.336394	.15252
.250	.99861	.9892	.9780	.9322	.9322	93.483206	.31570
.375	.99942	.9850	.9670	.9246	.9246	86.350813	.44887
.500	.99923	.9811	.9610	.9224	.9224	78.735278	.57788
.625	.99904	.9778	.9515	.9172	.9172	69.991269	.69352
.750	.99885	.9746	.9475	.9146	.9146	66.729493	.79852
.875	.99866	.9712	.9386	.9035	.9035	71.472905	.90460
1.000	.99847	.9690	.9329	.9121	.9121	42.267320	1.00000

(EXIT)

ELAPSED TIME IS .1000 HR , RELATIVE TIME IS 26.1375 96 INTEGR. STEPS  
 ALTITUDE IS 330000.00 FT , VELOCITY IS 16499.06 FT/SEC , ATM. TEMPERATURE IS 376.50 R

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, QSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT), INCIDENT INFRARED FLUX, QIRED = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION QTOT = .2110+05 BTU/HR , AERODYN. HEATING POWER, QCONV = .8142+04 BTU/HR  
 CONJ. FROM MANIFOLDS, CONDME = .2680+03 BTU/HR , ENERGY STORAGE RATE, STORG = -.1412+04 BTU/HR

RELATIVE TEMPERATURE OF FIN, T

DISTANCE NORMAL TO FLOW DIRECTION

Z	0	.0000	.25000	.50000	.75000	1.00000
.000	.7192	.9634	.9634	.9634	.9634	.9634
.125	.6096	.9257	.8948	.8876	.8852	.8852
.250	.5396	.9284	.9010	.8951	.8933	.8933
.375	.5289	.9188	.9048	.8955	.8902	.8885
.500	.5189	.9153	.9026	.8941	.8892	.8876
.625	.5029	.9096	.8984	.8910	.8867	.8854
.750	.5014	.9060	.8952	.8880	.8838	.8825
.875	.5144	.8959	.8848	.8772	.8728	.8714
1.000	.5421	.9050	.9050	.9050	.9050	.9050

FIGURE 10(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FI  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .164041+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = .112820+05 BTU/HR

PROTECT. LAYER TEMPERATURES

AXIAL DIST.	PRESSURE P	VELOCITY W	FLUID TEMP TF	WALL TEMP TW	PROTECT. LAYER TEMP TPL	ENTHALPY REJECTION PER UNIT TUBE LENGTH BTU/(HR FT)	REJECTION FRACTION OF TOTAL
.000	1.00000	1.0000	1.0000	.9634	.9634	73.935921	.0000
.125	.99981	.9936	.9850	.9257	.9257	119.899873	.15029
.250	.99961	.9885	.9766	.9264	.9264	101.527077	.31116
.375	.99942	.9840	.9651	.9186	.9186	94.038617	.44474
.500	.99923	.9798	.9582	.9153	.9153	86.704713	.57414
.625	.99904	.9760	.9482	.9096	.9096	78.045426	.69177
.750	.99885	.9726	.9429	.9060	.9060	74.709270	.79958
.875	.99866	.9690	.9337	.8959	.8959	76.516509	.90073
1.000	.99847	.9666	.9276	.9050	.9050	45.841238	1.00000



ELAPSED TIME IS .1250 HR , RELATIVE TIME IS 32.6718 108 INTEGR. STEPS 341.94 R  
 ALTITUDE IS 313506.32 FT , VELOCITY IS 23548.28 FT/SEC , ATM. TEMPERATURE IS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, QSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX, QREF = .3246+04 BTU/(HR FT), INCIDENT INFRARED FLUX, QIRD = .2000+01 BTU/(HR SQ FT)  
 PER UNIT AXIAL LENGTH, QTOT = .2112+05 BTU/HR , AERODYN. HEATING POWER, QCONV = .2673+05 BTU/HR  
 TOT. RADIANT REJECTION, CONDMF = .2058+03 BTU/HR , ENERGY STORAGE RATE, STORG = .1783+05 BTU/HR  
 COND. FROM MANIFOLDS,

AXIAL DIST. Z  
 RELATIVE RAD. HEAT REJECTION Q  
 RELATIVE TEMPERATURE OF FIN, T  
 DISTANCE NORMAL TO FLOW DIRECTION X

Z	Q	T	X
.000	.7023	.9583	.00000
.125	.6094	.9225	.25000
.250	.5488	.9222	.50000
.375	.5362	.9141	.75000
.500	.5244	.9098	1.00000
.625	.5097	.9035	
.750	.5053	.8991	
.875	.5052	.8949	
1.000	.5154	.8951	

FIGURE 10(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LRF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=10 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .156891+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = .120026+05 BTU/HR

AXIAL DIST. Z  
 PRESSURE P  
 VELOCITY W  
 FLUID TEMPERATURE TF  
 WALL TEMPERATURE TMI  
 PROTECT. LAYER TEMPERATURES TMP  
 ENTHALPY REJECTION PER UNIT TUBE LENGTH BTU/(HR.FT)  
 FRACTION OF TOTAL

Z	P	W	TF	TMI	TEMP	ENTHALPY REJECTION PER UNIT TUBE LENGTH BTU/(HR.FT)	FRACTION OF TOTAL
.000	1.00000	1.0000	1.0000	.9583	.9583	83.797215	.00000
.125	.9981	.9934	.9847	.9225	.9225	124.993712	.14961
.250	.99261	.9880	.9755	.9222	.9222	107.109274	.30702
.375	.99242	.9832	.9637	.9141	.9141	99.836724	.43992
.500	.9923	.9787	.9559	.9098	.9098	92.636692	.56679
.625	.99204	.9747	.9455	.9035	.9035	84.442323	.68700
.750	.99285	.9709	.9393	.8991	.8991	80.657710	.79626
.875	.99666	.9670	.9298	.8892	.8892	81.486121	.90382
1.000	.99847	.9642	.9229	.8951	.8951	55.786302	1.00000

(EXIT)

ELAPSED TIME IS .1270 HR , RELATIVE TIME IS 33.1959 110 INTEGR. STEPS 337.93 R  
 ALTITUDE IS 310980.00 FT , VELOCITY IS 24234.00 FT/SEC , ATM. TEMPERATURE IS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, QSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT), INCIDENT INFRARED FLUX, QIRED = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION ATOT = .2123+05 BTU/HR , AERODYN. HEATING POWER, QCONV = .3037+05 BTU/HR  
 CONJ. FROM MANIFOLDS, CONDMF = .1911+03 BTU/HR , ENERGY STORAGE RATE, STORG = .2135+05 BTU/HR

RELATIVE TEMPERATURE OF FIN, T

AXIAL DIST. RAD. HEAT REJECTION

DISTANCE NORMAL TO FLOW DIRECTION

Z	0	.00000	.25000	.50000	.75000	1.00000
.000	.7001	.9576	.9576	.9576	.9576	.9576
.125	.6115	.9226	.9122	.9032	.8973	.8951
.250	.5534	.9222	.9149	.9081	.9036	.9019
.375	.5406	.9140	.9081	.9022	.8982	.8967
.500	.5286	.9097	.9049	.8998	.8962	.8949
.625	.5145	.9034	.9000	.8960	.8930	.8919
.750	.5072	.8989	.8961	.8923	.8895	.8884
.875	.5066	.8991	.8962	.8923	.8793	.8782
1.000	.5127	.8941	.8941	.8941	.8941	.8941

FIGURE 10(A) CONCLUDED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER

\*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=I0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .156649+05 BTU/HR

INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = .120216+05 BTU/HR

AXIAL DIST.	PRESSURE P	VELOCITY W	FLUID T F	WALL T W	PROTECT. LAYER T M P	ENTHALPY REJECTION PER UNIT TUBE LENGTH OF BTU/(HR.FT)	FRACTION OF TOTAL
.000	1.00000	1.0000	1.0000	.9576	.9576	85.173653	.00000
.125	.99931	.9934	.9847	.9226	.9226	124.974236	.14983
.250	.99961	.9880	.9755	.9222	.9222	107.177940	.30672
.375	.99942	.9832	.9638	.9140	.9140	99.976870	.43945
.500	.99923	.9787	.9558	.9097	.9097	92.755548	.56813
.625	.99904	.9746	.9455	.9034	.9034	84.717443	.68622
.750	.99885	.9708	.9392	.8989	.8989	80.865383	.79553
.875	.99866	.9670	.9297	.8891	.8891	81.721740	.90304
1.000	.99847	.9641	.9227	.8941	.8941	57.510110	1.00000

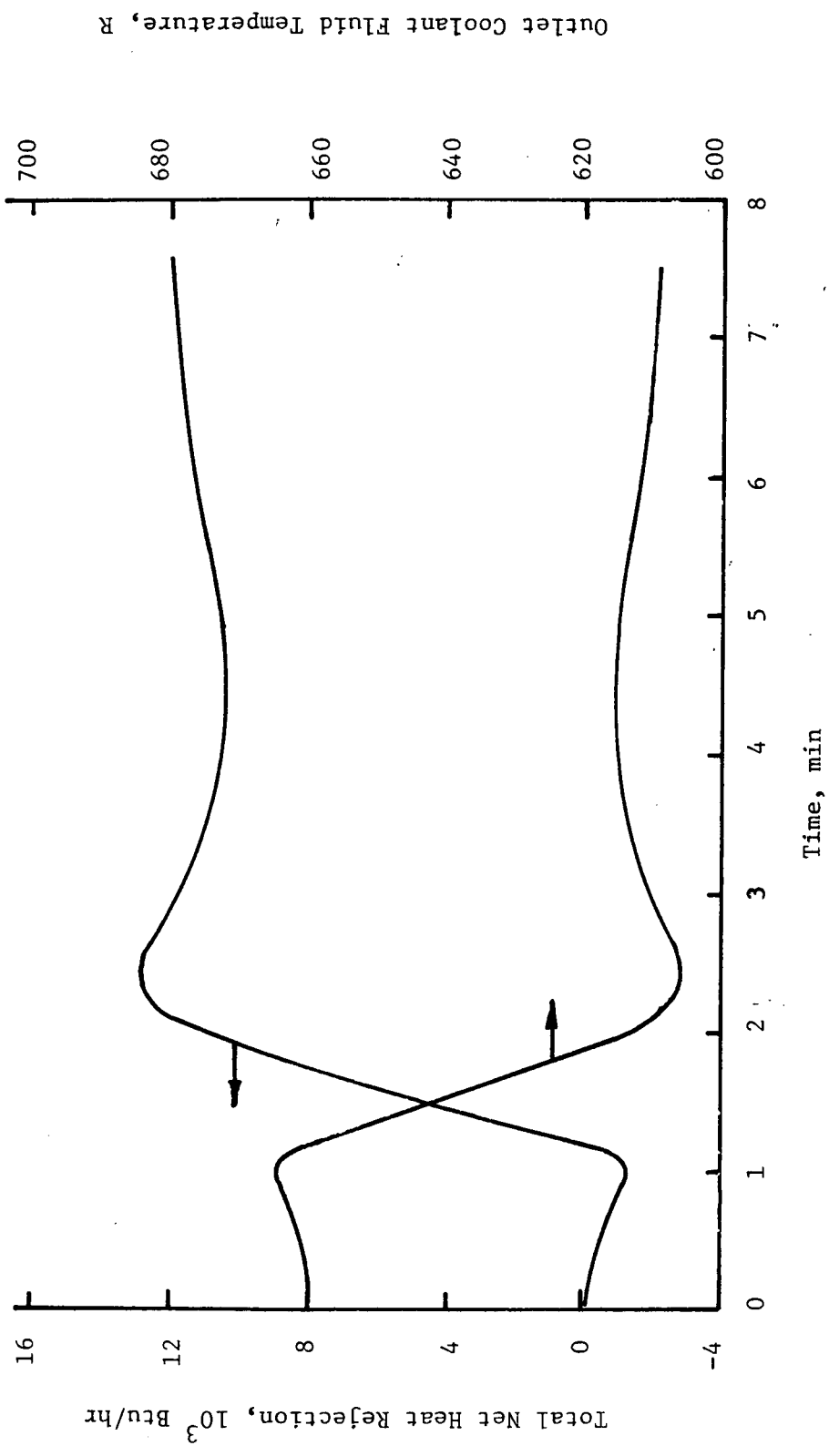


Fig. 10(b) System Response For Ascent--System Heat Rejection And Outlet Fluid Temperature

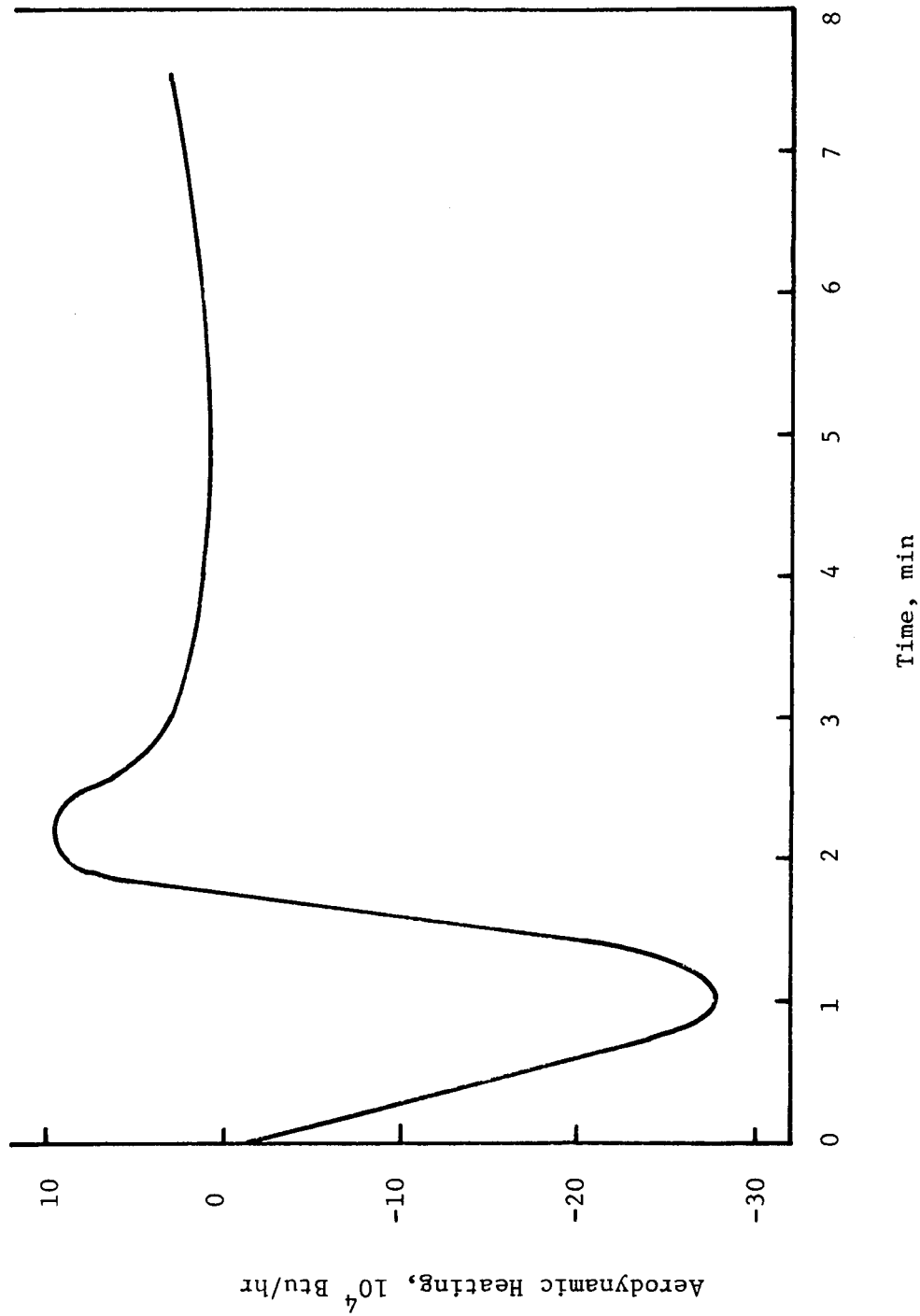


Fig. 11 Aerodynamic Heating For Ascent

b. Reentry

(i) Mission Description

The reentry altitude and velocity profiles used for the system output shown in Figures 13 and 14 are given in Figure 12.

Input parameters for reentry are identical to those given in the mission description for ascent except for  $IC\emptyset NV$  which is set equal to 2 to designate reentry. The system parameters for reentry are printed prior to the current system output shown in Figure 13a.

(ii) System Response

The response of the radiator system during reentry is summarized in Figures 13 and 14. Figure 13 (a) provides selected program output showing current system conditions at various times during the reentry phase of the mission. Figure 13 (b) is a plot of the total heat rejection from the radiator system and the temperature of the coolant fluid at the exit plane of the flow channel. Figure 14 is a graph of the aerodynamic heating of the system as a function of time during the reentry phase of the mission.

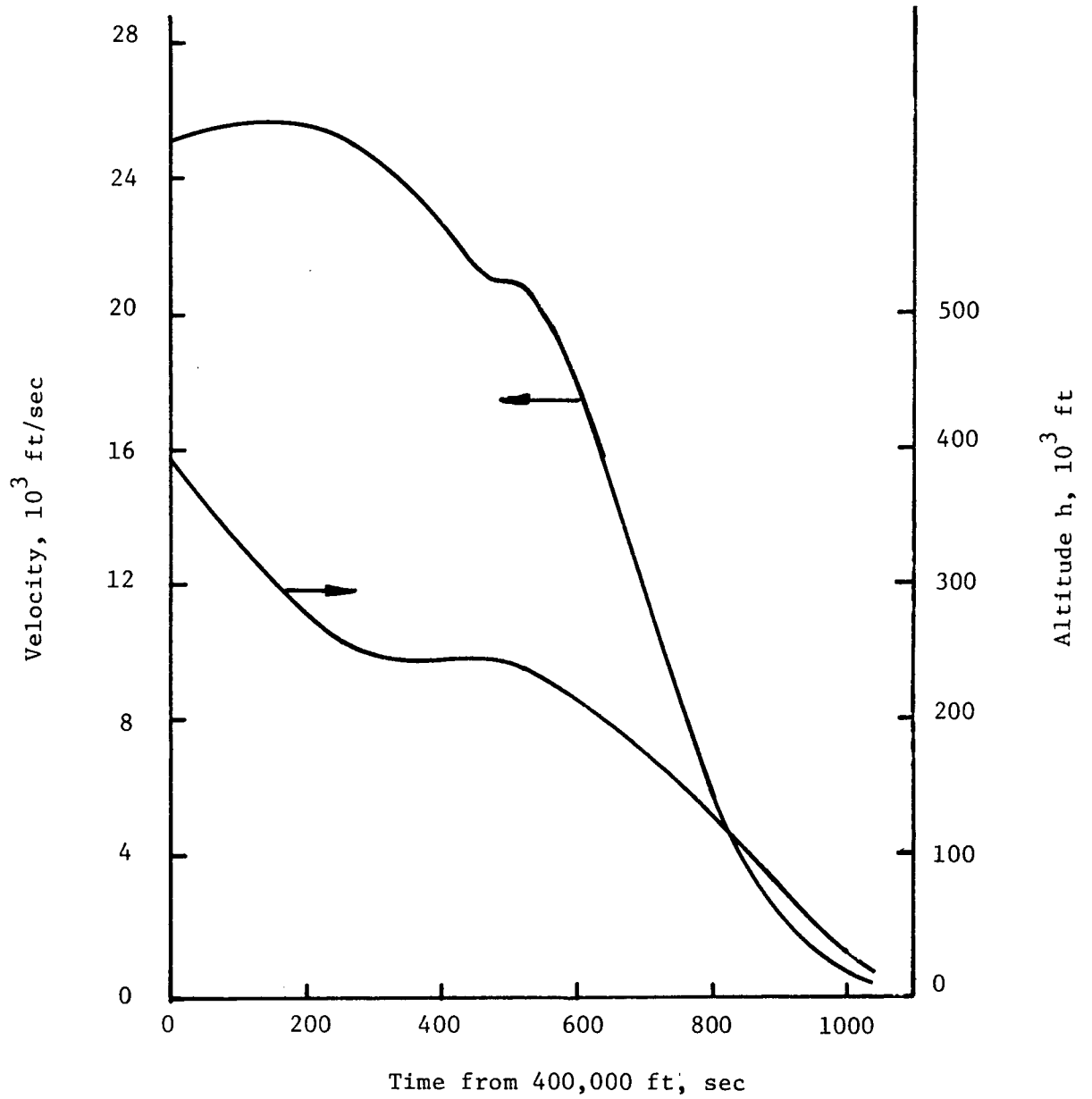


Fig. 12 Reentry Velocity and Altitude Profiles

```

SQMML =
ITAPE = +0
ICASE = +0
NTM = +4
IO = +0.00000000E+00
PHN = +0.00000000E+00

SEND
STUBE =
DITBI = +25000000E+00
SIBI = +1.00000000E+00
XL = +12000000E+02
RHOTBI = +1.685200E+03
MZ = +9
MRTBI = +3
NTBS = +10

SEND
$FLOW
MDOIT = +50000000E+03
IO = +55967000E+03
PO = +86400000E+04

```

FIGURE 13(A) SYSTEM RESPONSE FOR REENTRY PROGRAM OUTPUT

```

SEND
$FIN
NX = +5
SR00II = +50000000E-01
HFNT = +60000000E+01
STIPI = +50000000E-01
RHOFNI = +1.685200E+03
STAGX = +20000000E+02
VERTX = +10000000E+02

SEND
$PROJLR
NRMP = +3
RHOMPI = +1.685200E+03
RHOMET = +50000000E+00
VELM = +65600000E+05
TAU = +36500000E+04
PROB = +99000000E+00
ALPHA = +18800000E-09
BTA = +1.21300000E+01
GAMMA = +15000000E+01
PHI = +50000000E+00
THETA = +66666700E+00
AIK = +17500000E+01
AN = +10000000E+01

SEND
$MANIED
AMAN = +43000000E+02

SEND
$RUNOPT
MSTOTR = +2

```

88896

DTWRTE = .15667000E-01  
 IEND = .29400000E+00  
 ALIMIT = .50000000E-04  
 RLIMIT = .10000000E-04  
 TI = .65967000E+03  
 LIMINT = +25  
 NCONV = +2  
 LIT = +0  
 LFLD = +1  
 LTS = +0  
 SEND

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FIGURE 13(A) CONTINUED



INITIAL LINE CONDITIONS  
 \*\*\*\*\*  
 (ALL QUANTITIES ARE NORMALIZED)  
 \*\*\*\*\*

02

PT.NO.	POSITION Z	PRESSURE P	VELOCITY W	FLUID TEMPERATURE T	WALL TEMPERATURE TW
1	.000	1.000000	1.000000	1.000000	1.000000
2	.125	.999810	1.000000	1.000000	1.000000
3	.250	.999621	1.000001	1.000000	1.000000
4	.375	.999431	1.000001	1.000000	1.000000
5	.500	.999241	1.000001	1.000000	1.000000
6	.625	.999052	1.000001	1.000000	1.000000
7	.750	.998862	1.000001	1.000000	1.000000
8	.875	.998673	1.000002	1.000000	1.000000
9	1.000	.998483	1.000002	1.000000	1.000000

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INLET PRESSURE P0 = 8640.000 LBF/SQ.FT  
 REF. VELOCITY W0 = .87128 FT/SEC  
 REF. TEMPERATURE T00 = 659.670 R

REYNOLDS NO = .30017404  
 PRANDTL NO = 8.628066  
 DELTA = .001736  
 REL. PRESSURE IS = .391534404

Figure 13(A) CONTINUED

INIT. NUSSOLT NO. NU = .201856402  
 WALL BIOT NO. BI = .485314402

SYSTEM PARAMETERS  
 \*\*\*\*\*

TUBE LENGTH , XL = 12.000 FT  
 INTERNAL DIAMETER , IDTB = .250 IN  
 WALL THICKNESS , STB = .100 IN  
 MATERIAL = ALUMINUM  
 MASS (ALL TUBES) , MTB = 15.4480 LBM  
 NUMBER OF TUBES , NTBS = 10

FIN HEIGHT , HFN = 6.000 IN  
 THICKNESS AT FOOT , SROOT = .050 IN  
 THICKNESS AT TIP , STIP = .050 IN  
 MATERIAL = ALUMINUM  
 MASS (ALL FINES) , MFN = 84.2960 LBM  
 NO. OF FIN SIDES RADIATING = 1

COOLANT FLUID IS SILICONE OIL  
 MASS (IN ALL TUBES) , MFL = 19.1289 LBM

PROTECTION LAYER THICKNESS, SMP = .011 IN  
 MASS, MMP = 2.038 LBM  
 MATERIAL IS BERYLLIUM

TOTAL MASS (EXCL. MANIFLO.) MTOT = 120.9104 LBM

TOTAL AREA (SINGLE NORMAL PROJECTION) ,  
ATOT = 124,5000 SQ FT

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FIGURE 13(A) CONTINUED

ELAPSED TIME IS .0000 HR , RELATIVE TIME IS .0000 1 INTEGR. STEPS  
 ALTITUDE IS 39400.00 FT , VELOCITY IS 25400.00 FT/SEC , ATM. TEMPERATURE IS 661.41 R

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, QSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREE = .3246+04 BTU/(HR FT) , INCIDENT INFERRED FLUX, QIREN = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION GTOT = .3289+05 BTU/HR , AERODYN. HEATING POWER, QCONV = .6237+04 BTU/HR  
 CONVA. FROM MANIFOLDS, CONDME = .0000 BTU/HR , ENERGY STORAGE RATE, SIORG = -.2665+05 BTU/HR

AXIAL DIST.	RELATIVE RAD. HEAT REJECTION	RELATIVE TEMPERATURE OF FIN, T	DISTANCE NORMAL TO FLOW DIRECTION
Z	Q	T	X
.000	.8477	1.0000	.0000
.125	.8451	1.0000	.2500
.250	.8436	1.0000	.5000
.375	.8436	1.0000	.7500
.500	.8436	1.0000	1.0000
.625	.8430	1.0000	
.750	.8436	1.0000	
.875	.8451	1.0000	
1.000	.8477	1.0000	

FIGURE 13(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROJECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8650.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT IET0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .276866+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION OH = .500468-01 BTU/HR

AXIAL DIST.	PRESSURE P	VELOCITY W	FLUID TEMPERATURE T	WALL TEMPERATURE TW	PROTECT. LAYER TEMPERATURE TPL	ENTHALPY REJECTION PER UNIT TUBE LENGTH BIU/(HR.FT)	FRACTION OF TOTAL
.000	1.00000	1.0000	1.0000	1.0000	1.0000	.000000	.00000
.125	.99981	1.0000	1.0000	1.0000	1.0000	.000138	.02563
.250	.99962	1.0000	1.0000	1.0000	1.0000	.000242	.09431
.375	.99943	1.0000	1.0000	1.0000	1.0000	.000311	.19375
.500	.99924	1.0000	1.0000	1.0000	1.0000	.000380	.31625
.625	.99905	1.0000	1.0000	1.0000	1.0000	.000450	.46386
.750	.99886	1.0000	1.0000	1.0000	1.0000	.000484	.63147
.875	.99867	1.0000	1.0000	1.0000	1.0000	.000519	.80933
1.000	.99848	1.0000	1.0000	1.0000	1.0000	.000553	1.00000

ELAPSED TIME IS .0500 HR , RELATIVE TIME IS 13.0695 36 INTEGR. STEPS  
 ALTITUDE IS 292998.55 FT , VELOCITY IS 25799.98 EI/SEC , AIM. TEMPERATURE IS 325.17 R

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, QSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT), INCIDENT INFRARED FLUX, QIRED = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION QTOT = .2773+05 BTU/HR , AERODYN. HEATING POWER, QCONV = .5638+05 BTU/HR  
 CONDA. FROM MANIFOLDS, CONDMF = -.9819+02 BTU/HR , ENERGY STORAGE RATE, STORG = .3310+05 BTU/HR

AXIAL RELATIVE DIST. RAD. HEAT REJECTION Z g RELATIVE TEMPERATURE OF FIN, T DISTANCE NORMAL TO FLOW DIRECTION X

Z	g	T	X
.0000	.25000	.50000	.75000 1.00000
.125	.7159	.9625	.9625 .9625 .9625
.250	.7293	.9556	.9662 .9703 .9716 .9717
.375	.7242	.9527	.9643 .9690 .9707 .9709
.500	.7198	.9500	.9626 .9679 .9699 .9702
.625	.7259	.9478	.9610 .9668 .9691 .9694
.750	.7126	.9458	.9598 .9662 .9687 .9691
.875	.6797	.9441	.9589 .9657 .9684 .9690
1.000	.6273	.9345	.9345 .9345 .9345

FIGURE 13(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FI  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=10 HQ = .276867+05 BTU/HR INLET CURRENTLY HI = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .211363+05 BTU/HR TOT. REJECTION DH = .654839+04 BTU/HR

AXIAL PRESSURE VELOCITY FLUID WALL PROTECT. ENTHALPY REJECTION PER UNIT TUBE FRACTION OF DIST. P W W TF TEMPERATURES TWP TOTAL

AXIAL DIST.	P	W	W	TF	TEMPERATURES TWP	TOTAL
.000	1.00000	1.0000	1.0000	.9625	80.776558	.00000
.125	.99981	.9963	.9925	.9584	73.419073	.17633
.250	.99962	.9931	.9856	.9556	64.706697	.33361
.375	.99943	.9902	.9799	.9527	58.594118	.47323
.500	.99924	.9877	.9742	.9500	51.918591	.59957
.625	.99905	.9854	.9698	.9478	47.317073	.71200
.750	.99886	.9834	.9649	.9458	41.271317	.81355
.875	.99867	.9816	.9619	.9441	38.321868	.90322
1.000	.99848	.9794	.9578	.9345	50.009891	1.00000

ELAPSED TIME IS .1000 HR , RELATIVE TIME IS 26.1390 60 INTEGR. STEPS 362.54 R  
 ALTITUDE IS 245000.71 FT , VELOCITY IS 23799.82 FT/SEC , ATM. TEMPERATURE IS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, QSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FI), INCIDENT INFRARED FLUX, QIRKED = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION, QTOT = .5506+05 BTU/HR , AERODYN. HEATING POWER, QCONV = .1775+06 BTU/HR  
 COND. FROM MANIFOLDS, QCOND = .1573+03 BTU/HR , ENERGY STORAGE RATE, STORG = .1193+06 BTU/HR

AXIAL DIST. RELATIVE RAD. HEAT REJECTION  
 RELATIVE TEMPERATURE OF FIN, T  
 DISTANCE NORMAL TO FLOW DIRECTION

Z	0	.0000	.25000	.50000	.75000	1.00000
.000	.6299	.9354	.9354	.9354	.9354	.9354
.125	1.2481	1.0371	1.1389	1.2022	1.2359	1.2459
.250	1.6155	1.0278	1.1252	1.1849	1.2163	1.2254
.375	1.6081	1.0297	1.1254	1.1843	1.2153	1.2243
.500	1.5974	1.0254	1.1221	1.1816	1.2130	1.2222
.625	1.7098	1.0252	1.1178	1.1745	1.2043	1.2130
.750	1.5816	1.0219	1.1189	1.1789	1.2106	1.2200
.875	1.2127	1.0341	1.1332	1.1958	1.2296	1.2398
1.000	.6032	.9265	.9265	.9265	.9265	.9265

FIGURE 13(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROJECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBE/SQ.FI  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT I=I0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .307355+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = -.304877+04 BTU/HR

AXIAL DIST.	P	VELOCITY W	FLUID TEMPERATURE	WALL TEMPERATURE	PROTECT. LAYER TEMPERATURE	ENTHALPY REJECTION PER UNIT TUBE LENGTH BTU/(HR.FI)	FRACTION OF TOTAL
.000	1.00000	1.0000	1.0000	.9354	.9354	160.289324	.00000
.125	.99981	1.0040	1.0125	1.0371	1.0371	-61.108924	-.30040
.250	.99962	1.0060	1.0110	1.0278	1.0278	-41.625156	-.64967
.375	.99944	1.0070	1.0198	1.0297	1.0297	-24.422209	1.00147
.500	.99925	1.0081	1.0143	1.0254	1.0254	-27.635271	1.36675
.625	.99906	1.0079	1.0242	1.0252	1.0252	-2.552186	1.66864
.750	.99887	1.0084	1.0127	1.0219	1.0219	-23.023775	1.76029
.875	.99868	1.0095	1.0268	1.0341	1.0341	-18.077758	2.15741
1.000	.99850	1.0004	1.0171	.9265	.9265	224.826389	1.00000

ELAPSED TIME IS .1500 HR , RELATIVE TIME IS 39.2085 86 INTEGR. STEPS 384.00 R  
 ALTITUDE IS 234996.24 FT , VELOCITY IS 20199.73 FT/SEC , ATM. TEMPERATURE IS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION

REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, QSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT) , INCIDENT INFRARED FLUX, QIRED = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION GTOT = .6486+05 BTU/HR , AERODYN. HEATING POWER, QCONV = .1582+06 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .4866+03 BTU/HR , ENERGY STORAGE RATE, STORG = .8053+05 BTU/HR

AXIAL RELATIVE DIST. REJECTION DISTANCE NORMAL TO FLOW DIRECTION

Z	0	.0000	.25000	.50000	.75000	1.00000
.000	.7198	.9636	.9636	.9636	.9636	.9636
.125	1.4150	1.0926	1.1935	1.2578	1.2935	1.3045
.250	1.8367	1.0752	1.1646	1.2196	1.2491	1.2579
.375	1.8485	1.0866	1.1724	1.2249	1.2530	1.2614
.500	1.8592	1.0847	1.1701	1.2224	1.2504	1.2587
.625	1.9854	1.0895	1.1675	1.2141	1.2385	1.2456
.750	1.8776	1.0901	1.1758	1.2249	1.2523	1.2605
.875	1.5358	1.1232	1.2130	1.2699	1.3012	1.3108
1.000	.9599	1.0290	1.0290	1.0290	1.0290	1.0290

FIGURE 13(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBE/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT I=I0 HO = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .410317+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = -.133450+05 BTU/HR

AXIAL DIST.	RELATIVE REJECTION	VELOCITY W	TEMPERATURE T	WALL TEMPERATURE TW	PROTECT. LAYER TEMP	ENTHALPY REJECTION PER UNIT TUBE LENGTH BTU/(HR FT)	FRACTION OF TOTAL
.000	1.0000	1.0000	1.0000	.9636	.9636	99.048074	.00000
.125	.9992	1.0106	1.0293	1.0926	1.0926	-172.390614	.11927
.250	.9963	1.0168	1.0327	1.0752	1.0752	-115.685914	.35437
.375	.9944	1.0217	1.0514	1.0866	1.0866	-95.888686	.48609
.500	.9926	1.0265	1.0507	1.0847	1.0847	-92.660016	.62721
.625	.9907	1.0295	1.0694	1.0895	1.0895	-54.798841	.74575
.750	.9888	1.0332	1.0604	1.0901	1.0901	-80.840606	.83365
.875	.9869	1.0390	1.0841	1.1232	1.1232	-106.374774	.97867
1.000	.9850	1.0341	1.0833	1.0290	1.0290	147.963610	1.00000

(EXIT)

ELAPSED TIME IS .2000 HR , RELATIVE TIME IS 52.2780 112 INTEGR. STEPS  
 ALTITUDE IS 171993.57 FT , VELOCITY IS 10399.29 FT/SEC , ATM. TEMPERATURE IS 487.17 R

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REF. RADIANT HEAT FLUX, QREF = .3246\*04 BTU/(HR FT)<sup>2</sup> , INCIDENT SOLAR FLUX, QSOLR = .1000\*03 BTU/(HR SQ FT)  
 PER UNIT AXIAL LENGTH, QTOT = .7602\*05 BTU/HR , INCIDENT INFRARED FLUX, QIRSD = .2000\*01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION, CONDVE = .6873\*03 BTU/HR , AERODYN. HEATING POWER, QCONV = .1290\*06 BTU/HR  
 COND. FROM MANIFOLDS, CONDVE = .6873\*03 BTU/HR , ENERGY STORAGE RATE, STORG = .3334\*05 BTU/HR

AXIAL RELATIVE DIST. RAD. HEAT REJECTION FIN TEMPERATURE OF FIN, T DISTANCE NORMAL TO FLOW DIRECTION

Z	Q	T	X
.000	.7486	.9722	.50000
.125	1.6110	1.3218	.75000
.250	2.1982	1.2450	1.00000
.375	2.1632	1.1092	.75000
.500	2.1871	1.2119	1.00000
.625	2.3489	1.1246	.75000
.750	2.2311	1.1324	1.00000
.875	1.8337	1.1830	.75000
1.000	1.1597	1.0745	1.00000

FIGURE 13(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8650.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT I=10 H0 = .276867\*05 BTU/HR  
 EXIT CURRENTLY EI = .479690\*05 BTU/HR

AXIAL DIST. PRESSURE VELOCITY WALL TEMPERATURES PROTECT. LAYER ENTHALPY REJECTION PER UNIT TUBE LENGTH OF FRACTION TOTAL

AXIAL DIST.	P	W	TEMPERATURES	PROTECT. LAYER	ENTHALPY REJECTION PER UNIT TUBE LENGTH OF FRACTION TOTAL
.000	1.00000	1.0000	.9722	.9722	.00000
.125	.99982	1.0161	1.1342	1.1342	.13900
.250	.99963	1.0255	1.1092	1.1092	.35778
.375	.99945	1.0330	1.0764	1.0764	.48221
.500	.99926	1.0403	1.0770	1.1246	.61517
.625	.99907	1.0450	1.1030	1.1324	.72615
.750	.99889	1.0500	1.0923	1.1341	.81088
.875	.99870	1.0600	1.1250	1.1830	.94894
1.000	.99851	1.0553	1.1270	1.0745	1.00000

ELAPSED TIME IS 2500 HR , RELATIVE TIME IS 65.3475 138 INTEGR. STEPS 394.97 R  
 ALTITUDE IS 74990.18 FT. , VELOCITY IS 2199.64 FT/SEC. , ATM. TEMPERATURE IS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, GSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT) , INCIDENT INFRARED FLUX, GIRD = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION QTOT = .6239+05 BTU/HR , AERODYN. HEATING POWER, QCONV = -.2252+05 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .1221+09 BTU/HR , ENERGY STORAGE RATE, STORG = -.1037+06 BTU/HR

AXIAL RELATIVE RAD. HEAT REJECTION  
 DISTANCE NORMAL TO FLOW DIRECTION

Z	0	.00000	.25000	.50000	.75000	1.00000
.000	.7413	.9700	.9700	.9700	.9700	.9700
.125	1.2374	1.1293	1.1889	1.2386	1.2700	1.2810
.250	1.7043	1.1026	1.1505	1.1899	1.2143	1.2227
.375	1.7273	1.1216	1.1633	1.1992	1.2217	1.2295
.500	1.7503	1.1191	1.1605	1.1961	1.2196	1.2263
.625	1.8472	1.1295	1.1613	1.1902	1.2035	1.2149
.750	1.7916	1.1300	1.1681	1.2021	1.2237	1.2313
.875	1.5637	1.1812	1.2224	1.2614	1.2871	1.2965
1.000	1.2233	1.0887	1.0887	1.0887	1.0887	1.0887

FIGURE 13(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FI  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT I=10, H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .476853+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION OH = -.199986+05 BTU/HR

AXIAL DIST.	P	W	VELOCITY	FLUID TEMPERATURE	WALL TEMPERATURE	PROTECT. LAYER TEMPERATURE	ENTHALPY REJECTION PER UNIT TUBE LENGTH	FRACTION OF TOTAL
.000	1.0000	1.0000	1.0000	.9700	.9700	.9700	86.688877	.00000
.125	.99982	1.0156	1.0433	1.1293	1.1293	1.1293	-248.916988	.13119
.250	.99963	1.0244	1.0473	1.1026	1.1026	1.1026	-160.148674	.34140
.375	.99945	1.0316	1.0739	1.1216	1.1216	1.1216	-138.130276	.45934
.500	.99926	1.0386	1.0735	1.1191	1.1191	1.1191	-132.202616	.58956
.625	.99907	1.0434	1.0996	1.1295	1.1295	1.1295	-86.335973	.69966
.750	.99889	1.0491	1.0895	1.1300	1.1300	1.1300	-117.345061	.78642
.875	.99870	1.0585	1.1210	1.1812	1.1812	1.1812	-174.303867	.92608
1.000	.99851	1.0560	1.1255	1.0887	1.0887	1.0887	106.599895	1.00000



ELAPSED TIME IS .2667 HR , RELATIVE TIME IS 69.7040 148 INTEGR. STEPS 389.97 R  
 ALTITUDE IS 48989.54 FT , VELOCITY IS 1199.81 FT/SEC , ATM. TEMPERATURE IS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*  
 REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, QSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT) , INCIDENT INFRARED FLUX, QIRD = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION AT I=0, QTOT = .3745+05 BTU/HR , AERODYN. HEATING POWER, QCONV = -.1705+06 BTU/HR  
 COND. FROM MANIFOLDS, QCONDME = .1843+04 BTU/HR , ENERGY STORAGE RATE, STORG = -.2212+06 BTU/HR

AXIAL DIST.	RELATIVE RAD. HEAT REJECTION	RELATIVE TEMPERATURE OF FIN, T	DISTANCE NORMAL TO FLOW DIRECTION	
Z	Q	X	Y	T
.000	.8631	.0000	.25000	.50000
.125	.8816	1.0041	1.0041	1.0041
.250	.8987	1.0874	1.0392	1.0268
.375	.9144	1.0677	1.0163	1.0006
.500	.9291	1.0850	1.0277	1.0084
.625	.9486	1.0834	1.0259	1.0064
.750	.9553	1.0947	1.0311	1.0076
.875	1.1262	1.0951	1.0334	1.0117
1.000	1.4023	1.1242	1.0672	1.0398
(EXIT)			1.1242	1.1242

FIGURE 13(A) CONCLUDED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROJECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBE/SG.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT I=0, H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .427089+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = -.150223+05 BTU/HR

AXIAL DIST.	PROTECT. LAYER	WALL TEMPERATURE	FLUID TEMPERATURE	VELOCITY	PER UNIT TUBE LENGTH	ENTHALPY REJECTION FRACTION
P	IMP	IMI	TF	W	BTU/(HR.FT)	OF TOTAL
.000	1.0000	1.0041	1.0000	1.0000	-11.239679	.00000
.125	.9982	1.0874	1.0296	1.0103	-156.783590	.13991
.250	.9953	1.0677	1.0308	1.0160	-99.922624	.30879
.375	.9944	1.0850	1.0497	1.0212	-95.677561	.41267
.500	.9926	1.0834	1.0508	1.0262	-88.377534	.53137
.625	.9907	1.0683	1.0683	1.0306	-71.594559	.63305
.750	.9888	1.0348	1.0667	1.0348	-77.058642	.72233
.875	.9869	1.1330	1.0839	1.0421	-133.142769	.84743
1.000	.9850	1.1242	1.0955	1.0474	-78.039950	1.00000
(EXIT)						

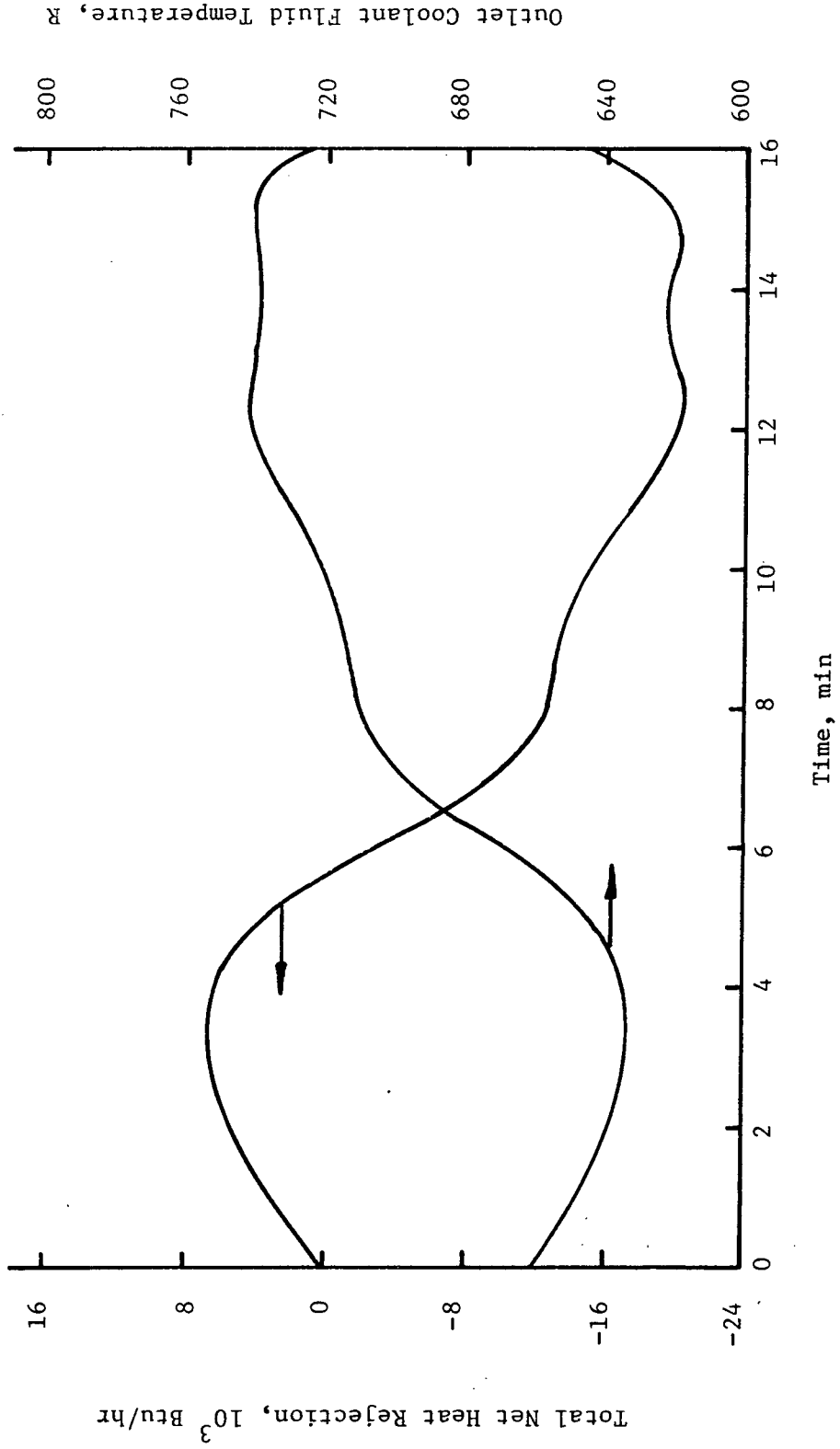


Fig. 13(b) System Response For Reentry-Heat Rejection And Outlet Fluid Temperature

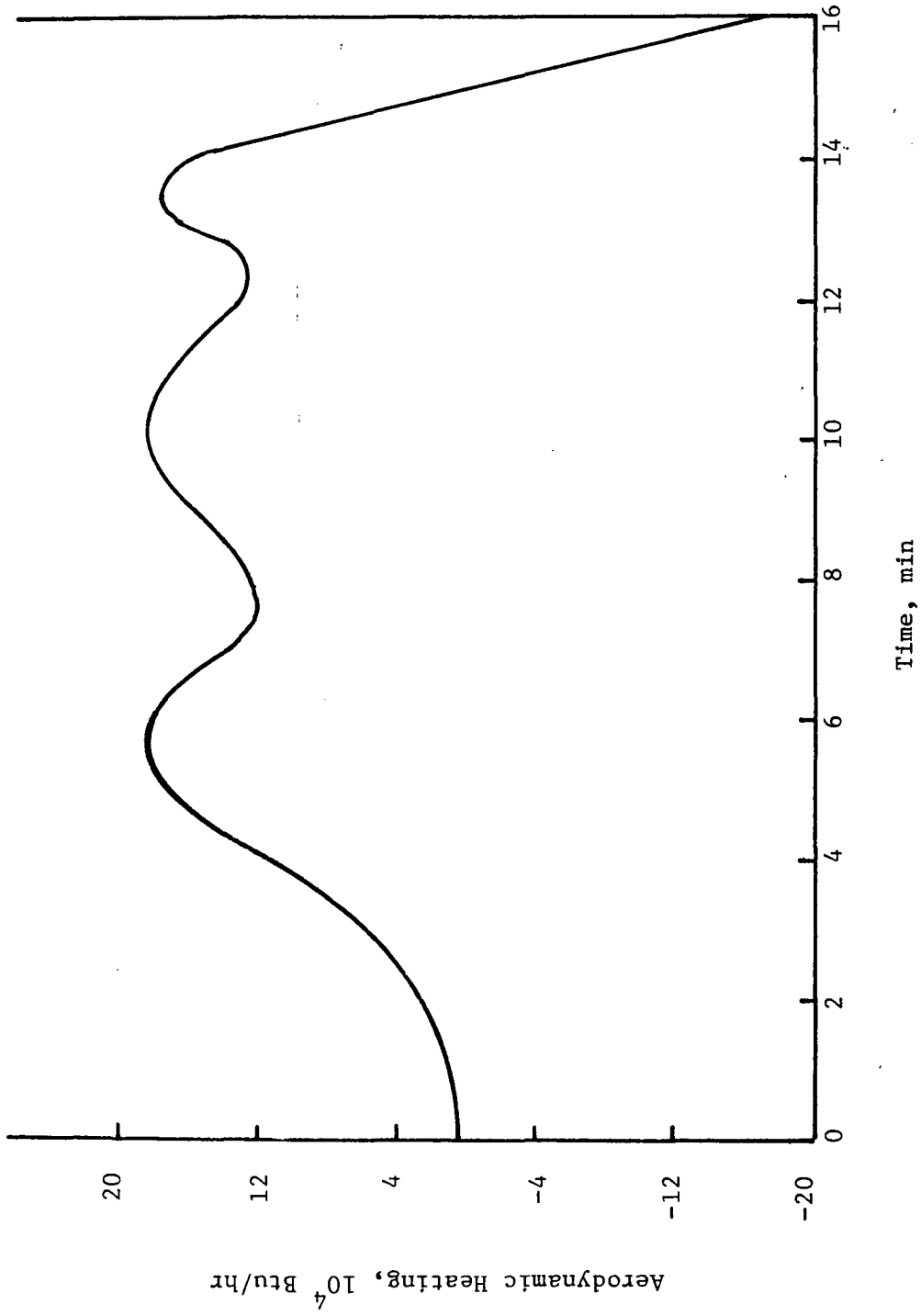


Fig.14 Aerodynamic Heating For Reentry

c. Orbital Mission

(i) Mission Description

Transient system performance is predicted in Figure 16 for a typical space base orbital mission. The mission is described by a 270 nautical mile circular orbit inclined at  $55^\circ$ . Incident flux values for solar irradiation, earth albedo and earth irradiation were supplied by a previously executed MRI program with the flux data being transferred from magnetic tape.

Incident radiant flux values for a typical fin element are plotted as a function of time into orbit in Figure 15. Other input parameters which specify system geometry, material properties and fluid flow conditions are identical to those used for the ascent and reentry phases that were previously reported in sections 5a and 5b.

(ii) System Response

The response of the radiator system during the first twenty minutes of the orbit is summarized in Figures 16(a) and 16(b). Input parameters and a selected number of output listings of current radiator conditions are shown in Figure 16(a). Figure 16(b) graphically summarizes the outlet coolant fluid temperature and the total net heat rejection from the radiator for the same period during orbit.

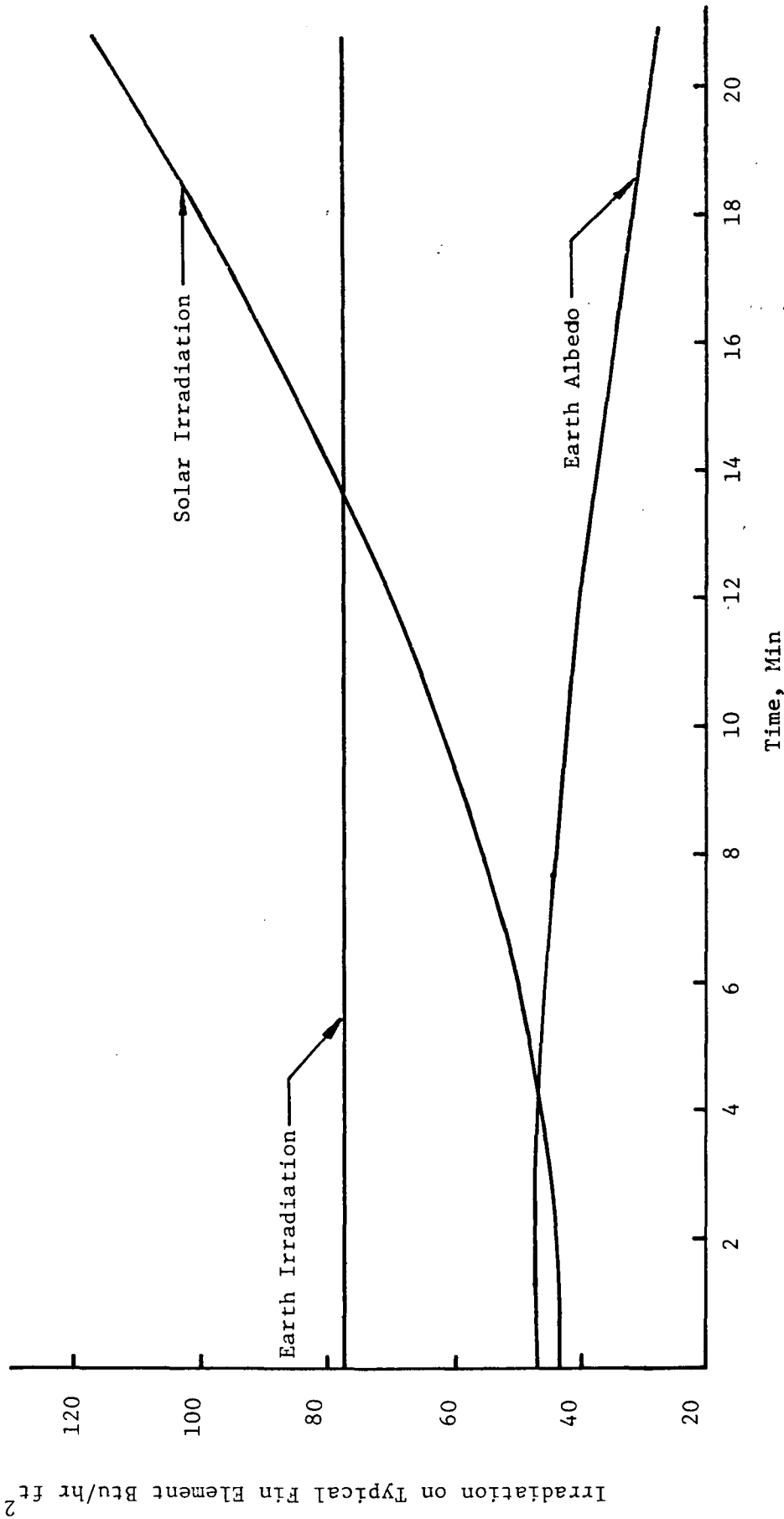


Fig. 15 Orbital Incident Fluxes

Irradiation on Typical Fin Element Btu/hr ft<sup>2</sup>

SGNML	=	+3
ITAPE	=	+1
ICASE	=	+73
NTM	=	+73
TO	=	.00000000E+00
PHN	=	.00000000E+00
SEND		
\$TUBE	=	.25000000E+00
DTBI	=	.10000000E+00
STBI	=	.12000000E+02
XL	=	.16859200E+03
RHOTBI	=	+9
KZ	=	+3
NKTB1	=	+3
NTBS	=	+10
SEND		
\$FLOW	=	.50000000E+03
M00TI	=	.65967000E+03
T0	=	.86400000E+04
P0	=	
SEND		
\$FIN	=	+5
NX	=	.50000000E-01
SRO0TI	=	.60000000E+01
RFNI	=	.50000000E-01
STIPI	=	.16859200E+03
RROFNI	=	.20000000E+02
STAGX	=	.10000000E+02
VERTX	=	
SEND		
\$PROTLR	=	+3
NKMP	=	.16859200E+03
RHOMPI	=	.50000000E+00
RHOMET	=	.65600000E+05
VELM	=	.36500000E+04
TAU	=	.99000000E+00
PROB	=	.18800000E-09
ALPHA	=	.12130000E+01
BTA	=	.15000000E+01
GAMMA	=	.50000000E+00
PHI	=	.66666700E+00
THETA	=	.17500000E+01
ATK	=	.10000000E+01
AN	=	
SEND		
\$MANIFD	=	.43000000E+02
AWAN	=	
SEND		
\$RUNOPT	=	+2
MSTOTR	=	

FIGURE 16(A) SYSTEM RESPONSE FOR ORBITAL MISSION  
PROGRAM OUTPUT

DTWRTE = .3500000E+01  
 TEND = .7000000E+00  
 ALIMIT = .5000000E-04  
 RLIMIT = .1000000E-04  
 T1 = .6596700E+03  
 LIMWRT = +25  
 NCONV = +0  
 LTT = +0  
 LFLD = +1  
 LTS = +0  
 SEND

FIGURE 16(A) CONTINUED

INITIAL LINE CONDITIONS  
 \*\*\*\*\*  
 (ALL QUANTITIES ARE NORMALIZED)

PT. NO.	POSITION Z	PRESSURE P	VELOCITY W	FLUID TEMPERATURE T	WALL TEMPERATURE TWI
1	.000	1.00000	1.00000	1.00000	1.00000
2	.125	.99910	1.00000	1.00000	1.00000
3	.250	.999621	1.000001	1.000000	1.000000
4	.375	.999431	1.000001	1.000000	1.000000
5	.500	.999241	1.000001	1.000000	1.000000
6	.625	.999052	1.000001	1.000000	1.000000
7	.750	.998862	1.000001	1.000000	1.000000
8	.875	.998673	1.000002	1.000000	1.000000
9	1.000	.998483	1.000002	1.000000	1.000000

INLET PRESSURE P0 = 8640.000 LBF/SQ.FT  
 REF. VELOCITY W0 = .87128 FT/SFC  
 REF. TEMPERATURE T00 = 659.670 R

REYNOLDS NO = .30017\*04  
 PRANDTL NO = 8.62R066  
 DELTA = .001736  
 REL.PRESSURE IS .391534\*04

Figure 16(A) CONTINUED

INIT. NUSSLET NO. NU = .201856\*02  
 WALL BYOT NO. BI = .485314-02

SYSTEM PARAMETERS  
 \*\*\*\*\*

TUBE LENGTH, XL = 12.000 FT  
 INTERNAL DIAMETER, DITB = .250 IN  
 WALL THICKNESS, STB = .100 IN  
 MATERIAL ALUMINUM  
 MASS (ALL TUBES), MTB = 15.4480 LBM  
 NUMBER OF TUBES, NTBS = 10  
 FIN HEIGHT, HFN = 6.000 IN  
 THICKNESS AT ROOT, SROOT = .050 IN  
 THICKNESS AT TIP, STIP = .050 IN  
 MATERIAL ALUMINUM  
 MASS (ALL FIN), MFN = 84.2960 LBM  
 NO. OF FIN SIDES RADIATING = 1

COOLANT FLUID IS SILYCOE OIL  
 MASS (IN ALL TUBES), MFL = 19.1289 LBM

PROTECTION LAYER THICKNESS, SMP = .011 IN  
 MASS, MMP = 2.038 LBM  
 MATERIAL IS BERYLLIUM

TOTAL MASS (EXCL. MANIFLD.) MTOT = 120.9104 LBM



TOTAL AREA (SINGLE NORMAL PROJECTION) :  
ATOT = 124,5000 SQ FT

72

FIGURE 16(A) CONTINUED

72

ELAPSED TIME IS .0000 HR  
 RELATIVE TIME IS .0000  
 \*\*IN ORBIT\*\*  
 1 INTEGR. STEPS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*  
 REFERENCE TEMPERATURE, T00 = 659.670 R  
 REF. RADIANT HEAT FLUX, QREF = .3246+04 BTU/(HR FT), INCIDENT SOLAR FLUX, QSOLR = .4326+03 BTU/(HR SQ FT)  
 PER UNIT AXIAL LENGTH, QREF = .2114+05 BTU/HR, INCIDENT INFRARED FLUX, QIRED = .1380+02 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION, QTOT = .0000 BTU/HR, AERODYN. HEATING POWER, QCONV = .0000 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .0000 BTU/HR, ENERGY STORAGE RATE, STORG = -.2114+05 BTU/HR

AXIAL RELATIVE TEMPERATURE OF FIN, T

AXIAL DIST.	RELATIVE REJECTION	RELATIVE TEMPERATURE OF FIN, T	DISTANCE NORMAL TO FLOW DIRECTION
Z	0	X	
.000	.5473	1.0000	.25000 .50000 .75000 1.00000
.125	.5439	1.0000	1.0000 1.0000 1.0000 1.0000
.250	.5418	1.0000	1.0000 1.0000 1.0000 1.0000
.375	.5418	1.0000	1.0000 1.0000 1.0000 1.0000
.500	.5418	1.0000	1.0000 1.0000 1.0000 1.0000
.625	.5412	1.0000	1.0000 1.0000 1.0000 1.0000
.750	.5418	1.0000	1.0000 1.0000 1.0000 1.0000
.875	.5439	1.0000	1.0000 1.0000 1.0000 1.0000
1.000	.5473	1.0000	1.0000 1.0000 1.0000 1.0000

(EXIT)

FIGURE 16(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .276866+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = .500486+01 BTU/HR

AXIAL DIST.	PRESSURE P	VELOCITY W	TEMPERATURES			PROTECT. LAYER TMP	ENTHALPY REJECTION PER UNIT TUBE LENGTH BTU/(HR FT)	FRACTION OF TOTAL
			FLUID TF	WALL TW	TEMP			
.000	1.00000	1.0000	1.0000	1.0000	1.0000	1.00000	.00000	.00000
.125	.99981	1.0000	1.0000	1.0000	1.0000	.000138	.02563	.000138
.250	.99962	1.0000	1.0000	1.0000	1.0000	.000242	.09431	.000242
.375	.99943	1.0000	1.0000	1.0000	1.0000	.000311	.19375	.000311
.500	.99924	1.0000	1.0000	1.0000	1.0000	.000380	.31625	.000380
.625	.99905	1.0000	1.0000	1.0000	1.0000	.000450	.46386	.000450
.750	.99886	1.0000	1.0000	1.0000	1.0000	.000484	.63147	.000484
.875	.99867	1.0000	1.0000	1.0000	1.0000	.000519	.80933	.000519
1.000	.99848	1.0000	1.0000	1.0000	1.0000	.000553	1.00000	.000553

(EXIT)

ELAPSED TIME IS .0700 HR  
 RELATIVE TIME IS 18.2969  
 \*\*IN ORBIT\*\*  
 44 INTEGR. STEPS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION

\*\*\*\*\*  
 REFERENCE TEMPERATURE, T00 = 659.670 R  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT)  
 TOT. RADIANT REJECTION QTOT = .1363+05 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .1923+03 BTU/HR  
 INCIDENT SOLAR FLUX, QSOLR = .4326+03 BTU/(HR SQ FT)  
 INCIDENT INFRARED FLUX, QIHED = .1313+02 BTU/(HR SQ FT)  
 AEROODYN. HEATING POWER, QCONV = .0000 BTU/HR  
 ENERGY STORAGE RATE, STORG = -.7238+04 BTU/HR

AXIAL DIST.	RELATIVE RAD. HEAT REJECTION	RELATIVE TEMPERATURE OF FIN, T	DISTANCE NORMAL TO FLOW DIRECTION
Z	O	X	
.000	.4812	.9816	.25000
.125	.3944	.9351	.50000
.250	.3403	.9368	.75000
.375	.3352	.9352	1.00000
.500	.3306	.9341	
.625	.3178	.9337	
.750	.3230	.9320	
.875	.3463	.9287	
1.000	.3877	.9538	

FIGURE 16(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER

\*\*\*\*\*  
 REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT I=I0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .218331+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = .625357+04 BTU/HR

AXIAL DIST.	PROTECT. LAYER	TEMPERATURE	WALL TWI	FLUID TF	VELOCITY W	ENTHALPY REJECTION PER UNIT TUBE LENGTH	FRACTION OF TOTAL
		TEMP				BTU/(HR FT)	
.000	1.0000	.9816	.9816	1.0000	1.0000	39.721291	.00000
.125	.9981	.9592	.9592	.9919	.9963	70.315785	.15369
.250	.9962	.9581	.9581	.9863	.9933	60.747243	.32489
.375	.9943	.9579	.9579	.9801	.9907	54.229073	.46043
.500	.9924	.9578	.9578	.9758	.9883	49.817427	.59891
.625	.9905	.9574	.9574	.9704	.9862	43.049000	.71844
.750	.9886	.9567	.9567	.9676	.9842	41.208278	.82419
.875	.9867	.9547	.9547	.9626	.9824	38.468452	.92672
1.000	.9848	.9538	.9538	.9600	.9816	13.438381	1.00000

ELAPSED TIME IS .1400 HR  
 RELATIVE TIME IS 36.5939  
 \*\*IN ORBIT\*\*  
 74 INTEGR. STEPS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT), INCIDENT SOLAR FLUX, QSOLR = .4326+03 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION QTOT = .1151+05 BTU/HR, AERODYN. HEATING POWER, GCONV = .1164+02 BTU/(HR SQ FT)  
 COND. FROM MANIFOLDS, CONDMF = .2601+03 BTU/HR, ENERGY STORAGE RATE, STORG = -.2599+04 BTU/HR

AXIAL DIST.	RELATIVE RAD. HEAT REJECTION	RELATIVE TEMPERATURE OF FIN, T	DISTANCE NORMAL TO FLOW DIRECTION X
0	.0000	.25000	.50000
.000	.4690	.9777	.9777
.125	.3587	.9273	.9150
.250	.2888	.9446	.9184
.375	.2791	.9392	.9247
.500	.2702	.9357	.9219
.625	.2516	.9316	.9191
.750	.2545	.9282	.9157
.875	.2787	.9212	.9089
1.000 (EXIT)	.3245	.9331	.9331

FIGURE 16(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT I=I0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .190346+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION QH = .865213+04 BTU/HR

AXIAL DIST.	FLUID PRESSURE P	VELOCITY W	TEMPERATURE T	WALL TEMPERATURE TW	PROTECT. LAYER TEMP	ENTHALPY REJECTION PER UNIT TUBE LENGTH	FRACTION OF TOTAL
0.000	1.00000	1.0000	1.0000	.9777	.9777	46.776074	.00000
.125	.98981	.9952	.9889	.9453	.9453	91.621171	.14379
.250	.99962	.9912	.9821	.9446	.9446	78.693769	.30749
.375	.99942	.9877	.9736	.9392	.9392	72.155310	.44290
.500	.99923	.9844	.9680	.9357	.9357	67.887362	.57417
.625	.99904	.9815	.9603	.9316	.9316	60.128966	.69518
.750	.99885	.9787	.9282	.9282	.9282	59.147844	.80515
.875	.99867	.9759	.9488	.9212	.9212	57.910906	.91549
1.000 (EXIT)	.99848	.9745	.9446	.9331	.9331	24.112062	1.00000

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*  
 REFERENCE TEMPERATURE, T00 = 659.670 R  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, GREF = .3246+04 BTU/(HR FT), INCIDENT SOLAR FLUX, QSOLR = .4326+03 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION QTOY = .1081+05 BTU/HR, AEROODYN. HEATING POWER, QCONV = .0000 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .2873+03 BTU/HR, ENERGY STORAGE RATE, STORG = -.9825+03 BTU/HR

AXIAL RELATIVE RAD. HEAT REJECTION

RELATIVE TEMPERATURE OF FIN, T

DIST. REJECTION

DISTANCE NORMAL TO FLOW DIRECTION

Z	0	.00000	.25000	.50000	.75000	1.00000
.000	.4659	.9763	.9763	.9763	.9763	.9763
.125	.3485	.9402	.9212	.9079	.9002	.8976
.250	.2736	.9401	.9236	.9037	.9036	.8976
.375	.2620	.9337	.9182	.9077	.9015	.8995
.500	.2510	.9297	.9149	.9048	.8989	.8970
.625	.2295	.9247	.9113	.9022	.8959	.8953
.750	.2309	.9206	.9071	.8979	.8925	.8908
.875	.2552	.9118	.8983	.8889	.8834	.8817
1.000	.3023	.9252	.9252	.9252	.9252	.9252

(EXIT)

FIGURE 16(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=T0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .181485+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = .953815+04 BTU/HR

AXIAL DIST.	PRESSURE P	VELOCITY W	TEMPERATURES			WALL TWI	PROTECT. LAYER TMP	ENTHALPY REJECTION PER UNIT TUBE LENGTH BTU/(HR FT)	FRACTION OF TOTAL
			TF	TI	TS				
.000	1.00000	1.0000	1.0000	.9763	.9763	.9763	49.157877	.00000	
.125	.99961	.9947	.9878	.9402	.9402	.9402	98.755130	.14067	
.250	.99962	.9905	.9807	.9401	.9401	.9401	84.352570	.30112	
.375	.99982	.9867	.9713	.9337	.9337	.9337	78.174984	.43370	
.500	.99923	.9831	.9654	.9297	.9297	.9297	74.254160	.56407	
.625	.99904	.9799	.9567	.9247	.9247	.9247	66.467727	.68532	
.750	.99885	.9768	.9525	.9206	.9206	.9206	66.269156	.79686	
.875	.99866	.9736	.9438	.9118	.9118	.9118	66.394603	.91070	
1.000	.99848	.9719	.9389	.9252	.9252	.9252	28.582312	1.00000	

(EXIT)

ELAPSED TIME IS .2800 HR  
 RELATIVE TIME IS 73.1878

136 INTEGR. STEPS  
 \*\*IN ORBIT\*\*

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT), INCIDENT SOLAR FLUX, QSOLR = .4326+03 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION QTOT = .1059+05 BTU/HR, AERODYN. HEATING POWER, GCONV = .0000 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .2989+03 BTU/HR, ENERGY STORAGE RATE, STORG = -.4030+03 BTU/HR

AXIAL RELATIVE REJECTION DIST. RAD. HEAT REJECTION

Z	RELATIVE TEMPERATURE OF FIN, T	DISTANCE NORMAL TO FLOW DIRECTION
.0000	.25000	.50000
.0000	.75000	1.00000
.0000	.9758	.9758
.125	.9758	.9758
.125	.9382	.9051
.250	.9382	.8946
.250	.9216	.9033
.375	.9315	.9012
.375	.9158	.8966
.500	.9274	.8958
.500	.9019	.8935
.625	.9220	.8917
.625	.9083	.8868
.750	.9176	.8868
.750	.9037	.8841
.875	.9230	.8783
.875	.9080	.8765
1.000	.9219	.9219
1.000	.9219	.9219

FIGURE 16(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LRF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FY/SEC  
 COOLANT POWER, INLET AT T=I0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .177966+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = .989009+04 BTU/HR

AXIAL DIST. PRESSURE VELOCITY WALL TEMPERATURES PROTECT. LAYER PER UNIT TUBE ENTHALPHY REJECTION FRACTION OF

AXIAL DIST.	P	W	TF	TWI	TEMPERATURES	TEMP	BTU/(HR FT)	TOTAL
.000	1.00000	1.0000	1.0000	.9758	.9758	.9758	50.129854	.00000
.125	.99981	.9946	.9873	.9382	.9382	.9382	101.470659	.13955
.250	.99961	.9903	.9802	.9384	.9384	.9384	86.378242	.29864
.375	.99942	.9863	.9704	.9315	.9315	.9315	80.419735	.42994
.500	.99923	.9826	.9644	.9274	.9274	.9274	76.643045	.55984
.625	.99904	.9793	.9553	.9220	.9220	.9220	68.930267	.68103
.750	.99885	.9760	.9510	.9176	.9176	.9176	69.137488	.79315
.875	.99866	.9727	.9418	.9080	.9080	.9080	70.083364	.90853
1.000	.99848	.9708	.9367	.9219	.9219	.9219	30.468460	1.00000

(EXIT)

ELAPSED TIME IS .3150 HR  
RELATIVE TIME IS 82.3363

151 INTEGR. STEPS  
\*\*IN ORBIT\*\*

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION

REFERENCE TEMPERATURE, T00 = 659.670 R  
REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT), INCIDENT SOLAR FLUX, QSOLR = .4326+03 BTU/(HR SQ FT)  
TOT. RADIANT REJECTION AT DT = .1056+05 BTU/HR, AERODYN. HEATING POWER, QCONV = .0000 BTU/HR  
CONDU. FROM MANIFOLDS, CONDMF = .3020+03 BTU/HR, ENERGY STORAGE RATE, STORG = -.2744+03 BTU/HR

AXIAL RELATIVE RAD. HEAT REJECTION  
DISTANCE NORMAL TO FLOW DIRECTION

Z	RELATIVE TEMPERATURE OF FIN, T	DISTANCE NORMAL TO FLOW DIRECTION X
.000	.9756	.75000 1.00000
.125	.9756	.9756 .9756
.250	.9377	.9180 .9044 .8937
.375	.9380	.9211 .9095 .9027 .9005
.500	.9309	.9151 .9042 .8979 .8958
.625	.9268	.9116 .9011 .8950 .8931
.750	.9213	.9074 .8980 .8925 .8908
.875	.9168	.9028 .8931 .8875 .8857
1.000	.9069	.8927 .8828 .8770 .8751
(EXIT)	.9211	.9211 .9211

FIGURE 16(A) CONCLUDED.

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER

REFERENCE TEMPERATURE, T00 = 659.670 R  
REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FT  
REFERENCE VELOCITY, W00 = .871 FT/SEC  
COOLANT POWER, INLET AT T=0 H0 = .276867+05 BTU/HR  
EXIT CURRENTLY E1 = .177034+05 BTU/HR  
INLET CURRENTLY HI = .276867+05 BTU/HR  
TOT. REJECTION DH = .998328+04 BTU/HR

AXIAL DIST.	PRESSURE P	VELOCITY W	FLUID TEMPERATURE TF	WALL TEMPERATURE TWI	PROTECT. LAYER TMP	ENTHALPY REJECTION PER UNIT TUBE LENGTH BTU/(HR FT)	FRACTION OF TOTAL
.000	1.00000	1.0000	1.0000	.9756	.9756	50.420656	.00000
.125	.9981	.9945	.9871	.9377	.9377	102.200647	.13930
.250	.9961	.9902	.9800	.9380	.9380	86.914325	.29805
.375	.9942	.9862	.9701	.9309	.9309	81.018184	.42902
.500	.9923	.9825	.9642	.9268	.9268	77.254283	.55879
.625	.9904	.9791	.9549	.9213	.9213	69.565755	.67993
.750	.9885	.9759	.9506	.9168	.9168	69.883926	.79218
.875	.9866	.9724	.9413	.9069	.9069	71.081735	.90794
1.000	.9848	.9706	.9361	.9211	.9211	30.978268	1.00000
(EXIT)							

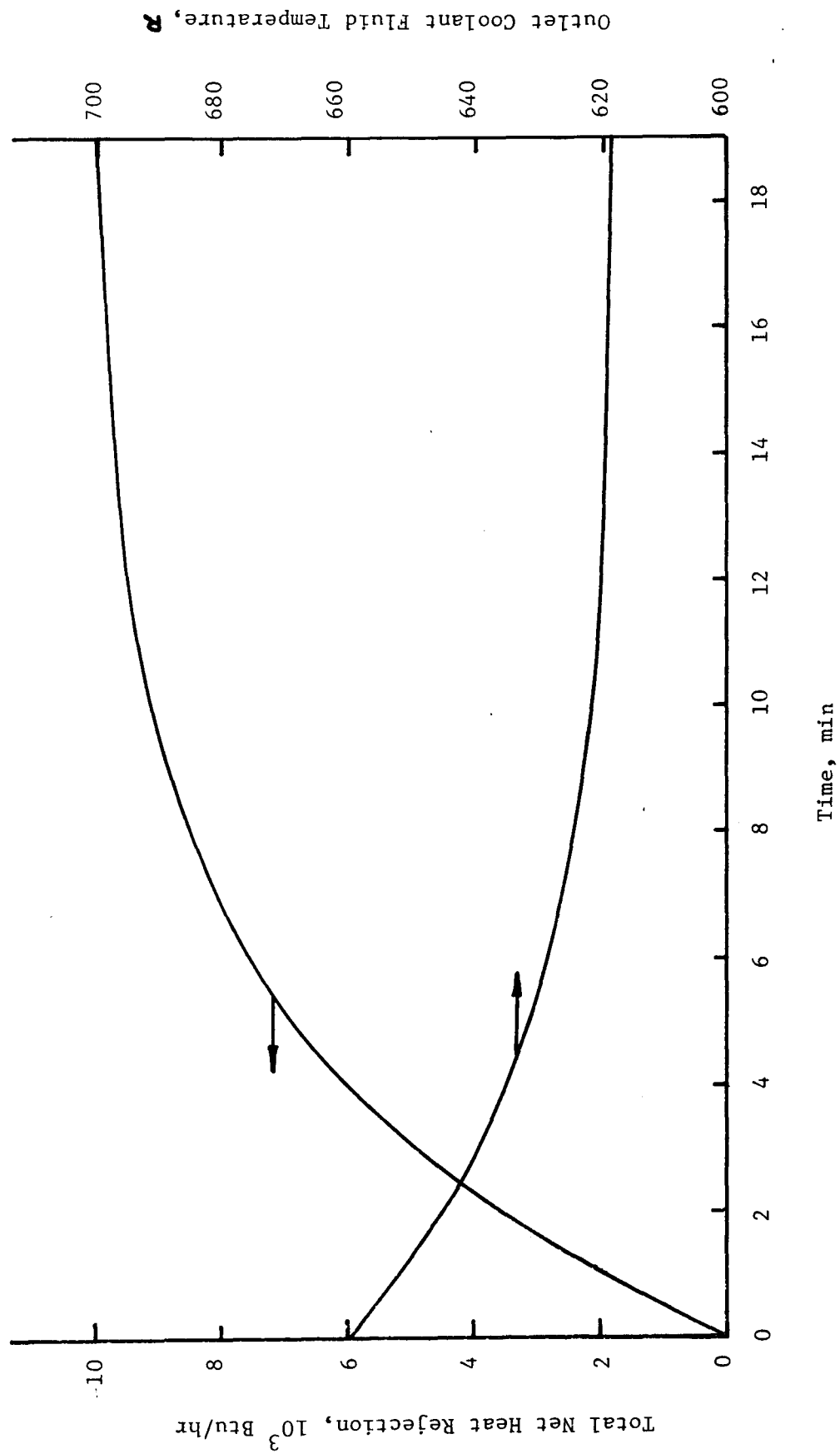


Fig. 16(b) System Response For Orbital Mission-Heat Rejection And Outlet Fluid Temperature



#### d. Steady State Environment

##### (i) Description of Environment

Program output has been produced for steady state system performance to a time invariant environmental conditions. The entire radiator system is initially assumed to be at a uniform temperature of 200°F (TI = 659.67R). The integration process is started and it proceeds until steady state operation of the system is achieved. The environmental conditions during the integration process remain constant. For the output of Figure 17, incident radiant flux values were read in from cards according to the format described in section A.3(d). These flux values were 100 Btu/hr. ft.<sup>2</sup> for solar irradiation, 2.0 Btu/hr. ft.<sup>2</sup> for earth albedo. There was assumed to be no infrared earth irradiation. There was no aerodynamic heating of the radiator panel. All other input parameters are the same as those used for program output during ascent (see section A.5(a)).

##### (ii) System Response

The response of the radiator system during the period leading to steady state operation is shown in Figure 17. Figure 17(a) shows the input parameters and a selected number of output listings of the system performance. Figure 17(b) graphically summarizes the resulting outlet coolant fluid temperature and the total heat rejection of the radiator system as the system approaches steady state operation.

```

$GNML =
ITAPE = +0
ICASE = +0
NTM = +4
TO = .00000000E+00
PHN = .00000000E+00

$END
$TUBE
GITBI = .25000000E+00
STBI = .10000000E+00
XL = .12000000E+02
RHOITBI = .16852000E+03 +5
M4 = +3
NHTBI = +3
NTBS = +10

```

FIGURE 17(A) SYSTEM RESPONSE FOR STEADY STATE ENVIRONMENT PROGRAM OUTPUT

```

$END
$FLOW
MLOTTI = .50000000E+03
TU = .65967000E+03
PO = .86400000E+04

$END
$FIN
NX = +5
SROOTI = .50000000E-01
PHNI = .60000000E+01
STIPI = .50000000E-01
RHOFINI = .16852000E+03
STAGX = .20000000E+02
VERTA = .10000000E+02

$END
$PROTLR
INRHP = +3
RHOMPI = .16852000E+03
KHONET = .50000000E+00
VELLM = .65600000E+05
TAU = .36500000E+04
PROB = .99000000E+00
ALPHA = .18800000E-09
BTA = .12130000E+01
GAMMA = .15000000E+01
PHI = .50000000E+00
THETA = .66666700E+00
ATK = .17500000E+01
AN = .10000000E+01

$END
$MANIFD
AMAN = .43000000E+02

$END
$RUNOPT
MSTOTR = +1

```

DTWRTE = .33333000E-01  
TEND = .00000000E+00  
ALIMIT = .50000000E-04  
RLIMIT = .10000000E-04  
TI = .65967000E+03  
LIMNT = +25  
NCONV = +0  
LTT = +0  
LFLO = +1  
LTS = +0

SEND

FIGURE 17(A) CONTINUED

INITIAL LINE CONDITIONS

\*\*\*\*\* (ALL QUANTITIES ARE NORMALIZED) \*\*\*\*\*

PT. NO.	POSITION Z	PRESSURE P	VELOCITY W	FLUID TEMPERATURE T	WALL TEMPERATURE TWI
1	.000	1.000000	1.000000	1.000000	1.000000
2	.250	.999621	1.000001	1.000000	1.000000
3	.500	.999241	1.000001	1.000000	1.000000
4	.750	.998862	1.000001	1.000000	1.000000
5	1.000	.998483	1.000002	1.000000	1.000000

INLET PRESSURE P0 = 8640.000 LBF/50.FT  
 REF. VELOCITY W0 = .87128 FT/SFC  
 REF. TEMPERATURE T00 = 659.670 R

REYNOLDS NO = .30017+04  
 PRANDTL NO = 8.628066  
 DELTA = .001736  
 REL. PRESSURE IS = .391534+04

FIGURE 17(A) CONTINUED

INIT. NUSSELT NO. NU = .201856+02  
 WALL BIOT NO. BI = .485314-02

SYSTEM PARAMETERS

\*\*\*\*\*

TUBE LENGTH, XL = 12.000 FT  
 INTERNAL DIAMETER, DITB = .250 IN  
 WALL THICKNESS, STB = .100 IN  
 MATERIAL = ALUMINUM  
 MASS (ALL TUBES), MTB = 15.4480 LBM  
 NUMBER OF TUBES, NTBS = 10

FIN HEIGHT, HFN = 6.000 IN  
 THICKNESS AT ROOT, SROOT = .050 IN  
 THICKNESS AT TIP, STIP = .050 IN  
 MATERIAL = ALUMINUM  
 MASS (ALL FIN), MFN = 84.2960 LBM  
 NO. OF FIN SIDES RADIATING = 1

COOLANT FLUID IS SILICONE OIL  
 MASS (IN ALL TUBES), MFL = 19.1289 LBM

PROTECTION LAYER THICKNESS, SMP = .011 IN  
 MASS, MMP = 2.038 LBM  
 MATERIAL IS BERYLLIUM

TOTAL MASS (EXCL. MANIFLD.) MTOT = 120.9104 LBM  
 TOTAL AREA (SINGLE NORMAL PROJECTION), ATOT = 124.5000 SQ FT

ELAPSED TIME IS .0000 HR  
 RELATIVE TIME IS .0000  
 \*\*IN ORBIT\*\*  
 1 INTEGR. STEPS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REF. RADIANT HEAT FLUX, GSOLR = .1000+03 BTU/(HR SQ FT)  
 PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT), INCIDENT INFRARED FLUX, GIREQ = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION, QTOT = .3288+05 BTU/HR, AERODYN. HEATING POWER, GCONV = .0000 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .0000 BTU/HR, ENERGY STORAGE RATE, STORG = -.3288+05 BTU/HR

AXIAL DIST.	RELATIVE RAD. HEAT REJECTION	RELATIVE TEMPERATURE OF FIN, T	DISTANCE NORMAL TO FLOW DIRECTION
Z	Q	X	X
.000	.8477	1.0000	.25000
.250	.8436	1.0000	.50000
.500	.8436	1.0000	.75000
.750	.8436	1.0000	1.00000
1.000	.8477	1.0000	1.00000

FIGURE 17(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=10, H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .276866+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = .498047-01 BTU/HR

AXIAL DIST.	PROTECT. LAYER TEMPERATURE	WALL TEMPERATURE	FLUID VELOCITY	PROTECT. LAYER THICKNESS	ENTHALPY REJECTION PER UNIT TUBE LENGTH	FRACTION OF TOTAL
Z	T	TW	W	TF	BTU/(HR FT)	
.000	1.0000	1.0000	1.0000	1.0000	.00000	.00000
.250	.9962	1.0000	1.0000	1.0000	.000277	.10085
.500	.9924	1.0000	1.0000	1.0000	.000415	.33176
.750	.9886	1.0000	1.0000	1.0000	.000519	.63619
1.000	.9848	1.0000	1.0000	1.0000	.000588	1.00000

ELAPSED TIME IS .1000 HR  
 RELATIVE TIME IS 26.1382  
 \*\*IN ORBIT\*\*  
 52 INTEGR. STEPS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REF. RADIANT HEAT FLUX, Q50LR = .1000\*03 BTU/(HR SQ FT)  
 PER UNIT AXIAL LENGTH, QREF = .3246\*04 BTU/(HR FT), Q1RED = .2000\*01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION, QTOT = .1577\*05 BTU/HR, AERODYN. HEATING POWER, QCONV = .0000 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .1545\*03 BTU/HR, ENERGY STORAGE RATE, STORG = -.7621\*04 BTU/HR

AXIAL DIST.	RELATIVE REJECTION	RELATIVE TEMPERATURE OF FIN, T	DISTANCE NORMAL TO FLOW DIRECTION
Z	0	X	
.0000	.25000	.50000	.75000 1.00000
.000	.9669	.9669	.9669
.250	.9203	.8851	.8770 .8744
.500	.8808	.8493	.8777 .8704 .8681
.750	.8415	.8174	.8709 .8647 .8628
1.000	.8076	.8070	.9070 .9070

FIGURE 17(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=10 H0 = .276867\*05 BTU/HR  
 EXIT CURRENTLY EI = .156871\*05 BTU/HR  
 INLET CURRENTLY HI = .276867\*05 BTU/HR  
 TOT. REJECTION DH = .119996\*05 BTU/HR

AXIAL DIST.	PRESSURE P	VELOCITY W	TEMPERATURE T	WALL TEMP	PROTECT. LAYER TEMPS	ENTHALPY REJECTION PER UNIT TUBE LENGTH	FRACTION OF TOTAL
			TF	TWI	TMP	BTU/(HR.FT)	
.000	1.00000	1.0000	1.0000	.9669	.9669	69.442389	.00000
.250	.99962	.9804	.9740	.9203	.9203	112.865777	.27736
.500	.99925	.9788	.9559	.9091	.9091	98.273507	.58257
.750	.99886	.9709	.9354	.8974	.8974	79.855536	.82653
1.000	.99848	.9668	.9238	.9070	.9070	35.347144	1.00000

ELAPSED TIME IS .2000 HR  
 RELATIVE TIME IS 52.2765 \*\*IN ORBIT\*\* 79 INTEGR. STEPS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REF. RADIANT HEAT FLUX, QREF = .3246+04 BTU/(HR FT), INCIDENT SOLAR FLUX, QSOLR = .1000+03 BTU/(HR SQ FT)  
 PER UNIT AXIAL LENGTH, QREF = .1740+05 BTU/HR, INCIDENT INFRARED FLUX, QIRED = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION, QTOT = .1896+03 BTU/HR, AERODYN. HEATING POWER, GCONV = .0000 BTU/HR  
 CONJ. FROM MANIFOLDS, CONDMF = .1896+03 BTU/HR, ENERGY STORAGE RATE, STORG = -.2265+04 BTU/HR

AXIAL DIST. RELATIVE RAD. HEAT REJECTION  
 Z Q

AXIAL DIST.	RELATIVE RAD. HEAT REJECTION Q	RELATIVE TEMPERATURE OF FIN, T	DISTANCE NORMAL TO FLOW DIRECTION X
.000	.7183	.9632	.50000
.250	.4485	.8798	.75000
.500	.4166	.8657	1.00000
.750	.3838	.8501	
1.000	.4785	.8810	

FIGURE 17(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LRF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=I0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .127453+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION OH = .14941+05 BTU/HR

AXIAL DIST.	FLUID TEMPERATURES	PROTECT. LAYER TEMPERATURES	ENTHALPY REJECTION PER UNIT TUBE LENGTH	FRACTION OF TOTAL
	P	W	BTU/(HR FT)	
.000	1.0000	1.0000	75.726097	.00000
.250	.99961	.9863	133.561531	.25844
.500	.9923	.9745	122.299504	.55811
.750	.99886	.9641	104.992007	.81310
1.000	.99848	.9585	49.425550	1.00000

ELAPSED TIME IS .3000 HR  
 RELATIVE TIME IS 78.4147  
 \*\*IN ORBIT\*\*  
 104 INTEGR. STEPS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, QSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT), INCIDENT INFRARED FLUX, QIRDN = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION QTOT = .1685+05 BTU/HR , AERODYN. HEATING POWER, QCONV = .0000 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .1986+03 BTU/HR , ENERGY STORAGE RATE, STORG = -.9773+03 BTU/HR

AXIAL RELATIVE DIST. RAD. HEAT REJECTION

Z	g	RELATIVE TEMPERATURE OF FIN, T	DISTANCE NORMAL TO FLOW DIRECTION	X
	.0000	.25000	.50000	.75000 1.00000
	.000	.7153	.9624	.9624 .9624
	.250	.4374	.9005	.8755 .8582 .8481 .8448
	.500	.4025	.8836	.8600 .8437 .8342 .8311
	.750	.3645	.8625	.8417 .8273 .8189 .8162
	1.000	.3617	.8743	.8743 .8743

FIGURE 17(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .120168+05 BTU/HR  
 INLET CURRENTLY HI = .270867+05 BTU/HR  
 TOT. REJECTION DH = .156659+05 BTU/HR

AXIAL DIST.	PRESSURE P	VELOCITY W	FLUID TEMPERATURE TF	WALL TEMPERATURE TWI	PROTECT. LAYER TEMPERATURE TMP	ENTHALPY REJECTION PER UNIT TUBE LENGTH BTU/(HR FT)	FRACTION OF TOTAL
.000	1.00000	1.0000	1.0000	.9624	.9624	76.930037	.00000
.250	.99961	.9858	.9678	.9005	.9005	137.760096	.25316
.500	.99923	.9735	.9462	.8836	.8836	128.067226	.55047
.750	.99886	.9625	.9173	.8625	.8625	112.156114	.80811
1.000	.99848	.9565	.9003	.8743	.8743	53.326069	1.00000



ELAPSED TIME IS .4000 HR  
 RELATIVE TIME IS 104.5529

\*\*IN ORBIT\*\*

130 INTEGR. STEPS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION

\*\*\*\*\*  
 REFERENCE TEMPERATURE, T00 = 659.670 R  
 REF. RADIANT HEAT FLUX  
 PER UNIT AXIAL LENGTH, QREF = .3246+04 BTU/(HR FT), INCIDENT SOLAR FLUX, QSOLR = .1000+03 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION, QTOT = .1671+05 BTU/HR, INCIDENT INFRARED FLUX, QIRED = .2000+01 BTU/(HR SQ FT)  
 CONJ. FROM MANIFOLDS, CONDMF = .2010+03 BTU/HR, AERODYN. HEATING POWER, QCONV = .0000 BTU/HR  
 ENERGY STORAGE RATE, STOR6 = -.6569+03 BTU/HR

AXIAL DIST. RELATIVE RAD. HEAT REJECTION  
 DISTANCE NORMAL TO FLOW DIRECTION

Z	0	.0000	.25000	.50000	.75000	1.00000
.000	.7152	.9622	.9622	.9622	.9622	.9622
.250	.4378	.8745	.8571	.8409	.8236	.8064
.500	.3993	.8825	.8822	.8822	.8822	.8822
.750	.3595	.8607	.8395	.8249	.8163	.8135
1.000	.4574	.8725	.8725	.8725	.8725	.8725

FIGURE 17(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER

\*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE P-PRESSURE, P00 = 8640.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=TO H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .118339+05 BTU/HR

INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = .158528+05 BTU/HR

AXIAL DIST.	PROTECT. LAYER	FLUID TEMPERATURE	WALL TEMPERATURE	ENTHALPY REJECTION PER UNIT TUBE LENGTH	FRACTION OF TOTAL
P	TF	W	TI	BTU/(HR.FT)	
.000	1.0000	1.0000	.9622	77.176887	.00000
.250	.9961	.9857	.8997	138.666939	.25172
.500	.9923	.9733	.8825	129.416553	.54823
.750	.9986	.9621	.8607	114.102509	.80655
1.000	.9988	.9560	.8725	54.379859	1.00000

ELAPSED TIME IS .5000 HR  
 RELATIVE TIME IS 130.6912  
 \*\*IN ORBIT\*\* 157 INTEGR. STEPS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, GREF = .3246+04 BTU/(HR FT), INCIDENT SOLAR FLUX, GSOLR = .1000+03 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION QTOT = .1668+05 BTU/HR, AERODYN. HEATING POWER, GCONV = .0000 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .2017+03 BTU/HR, ENERGY STORAGE RATE, STORG = -.5757+03 BTU/HR

AXIAL RELATIVE DIST. RAD. HEAT REJECTION  
 RELATIVE TEMPERATURE OF FIN, T  
 DISTANCE NORMAL TO FLOW DIRECTION

Z	0	.0000	.25000	.50000	.75000	1.00000
.000	.7150	.9622	.9622	.9622	.9622	.9622
.250	.4342	.8994	.8742	.8568	.8466	.8433
.500	.3985	.8822	.8584	.8419	.8322	.8290
.750	.3582	.8602	.8390	.8243	.8156	.8128
1.000	.4564	.8721	.8721	.8721	.8721	.8721

FIGURE 1Z(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=T0 H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .117874+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION OH = .158993+05 BTU/HR

AXIAL DIST.	PRESSURE P	VELOCITY W	FLUID TEMPERATURES			PROTECT. LAYER TEMPERATURES		ENTHALPY REJECTION	
			TF	TW	TI	TP	PER UNIT TUBE LENGTH	FRACTION OF TOTAL	
.000	1.00000	1.00000	1.0000	.9622	.9622	.9622	77.232511	.00000	
.250	.99961	.9857	.9674	.8994	.8994	.8994	138.877901	.25135	
.500	.99923	.9732	.9457	.8822	.8822	.8822	129.726835	.54762	
.750	.99886	.9620	.9163	.8602	.8602	.8602	114.610315	.80810	
1.000	.99848	.9558	.8989	.8721	.8721	.8721	54.654328	1.00000	

ELAPSED TIME IS .6000 HR 184 INTEG. STEPS  
 RELATIVE TIME IS 156.8294 \*\*IN ORBIT\*\*

90

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, OSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX, QREF = .3246+04 BTU/(HR FT), INCIDENT INFRARED FLUX, OIRRED = .2000+01 BTU/(HR SQ FT)  
 PER UNIT AXIAL LENGTH, QREF = .1667+05 BTU/HR , AERODYN. HEATING POWER, QCONV = .0000 BTU/HR  
 TOT. RADIANT REJECTION, QTOT = .2018+03 BTU/HR , ENERGY STORAGE RATE, STORG = -.5547+03 BTU/HR  
 COND. FROM MANIFOLDS, CONDMF = .8720 .8720 .8720 .8720 .8720 .8720 .8720 .8720 .8720 .8720

AXIAL DIST.	RELATIVE TEMPERATURE OF FIN, T	RELATIVE TEMPERATURE OF FIN, T	DISTANCE NORMAL TO FLOW DIRECTION
0	.0000	.5000	.7500 1.0000
.000	.7150	.9622	.9622 .9622 .9622
.250	.4340	.8994	.8741 .8567 .8465 .8432
.500	.3983	.8822	.8583 .8418 .8321 .8289
.750	.3579	.8601	.8388 .8241 .8155 .8127
1.000	.4561	.8720	.8720 .8720 .8720 .8720

FIGURE 17(A) CONTINUED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LBF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=TO H0 = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .117753+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = .159114+05 BTU/HR

AXIAL DIST.	PRESSURE P	VELOCITY W	FLUID TEMPERATURES	WALL TEMPERATURES	PROTECT. LAYER TEMPERATURES	ENTHALPY REJECTION PER UNIT TUBE LENGTH	FRACTION OF TOTAL
.000	1.0000	1.0000	1.0000	.9622	.9622	77.246378	.00000
.250	.99961	.9857	.9674	.8994	.8994	138.930130	.25127
.500	.99923	.9732	.9457	.8822	.8822	129.798357	.54747
.750	.99886	.9620	.9162	.8601	.8601	114.739276	.80599
1.000	.99848	.9558	.8988	.8720	.8720	54.722584	1.00000

(EXIT)

ELAPSED TIME IS .6183 HR  
 RELATIVE TIME IS 161.6107

\*\*IN ORBIT\*\*

188 INTEGR. STEPS

FIN TEMPERATURE DISTRIBUTION AND RADIANT HEAT REJECTION  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R , INCIDENT SOLAR FLUX, GSOLR = .1000+03 BTU/(HR SQ FT)  
 REF. RADIANT HEAT FLUX PER UNIT AXIAL LENGTH, GREF = .3246+04 BTU/(HR FT), INCIDENT INFRARED FLUX, GIRED = .2000+01 BTU/(HR SQ FT)  
 TOT. RADIANT REJECTION, GTOT = .1667+05 BTU/HR , AERODYN. HEATING POWER, GCONV = .0000 BTU/HR  
 CONJ. FROM MANIFOLDS, CONDMF = .2019+03 BTU/HR , ENERGY STORAGE RATE, STORG = -.5533+03 BTU/HR

AXIAL RELATIVE RAD. HEAT REJECTION

AXIAL DIST.	RELATIVE RAD. HEAT REJECTION	RELATIVE TEMPERATURE OF FIN, T
0	.0000	.75000 1.00000
.000	.9622	.9622 .9622 .9622
.250	.9994	.8741 .9567 .8432
.500	.9822	.8583 .9418 .8321 .8289
.750	.8600	.8388 .8241 .8155 .8126
1.000	.8720	.8720 .8720 .8720
(EXIT)		

FIGURE 17(A) CONCLUDED

FLUID PROPERTIES AND SURFACE TEMPERATURE OF METEOROID PROTECTION LAYER  
 \*\*\*\*\*

REFERENCE TEMPERATURE, T00 = 659.670 R  
 REFERENCE PRESSURE, P00 = 8640.000 LRF/SQ.FT  
 REFERENCE VELOCITY, W00 = .871 FT/SEC  
 COOLANT POWER, INLET AT T=0 HO = .276867+05 BTU/HR  
 EXIT CURRENTLY EI = .117746+05 BTU/HR  
 INLET CURRENTLY HI = .276867+05 BTU/HR  
 TOT. REJECTION DH = .159121+05 BTU/HR

AXIAL DIST. PRESSURE VELOCITY W

AXIAL DIST.	PRESSURE P	VELOCITY W	TEMPERATURE T	PROTECT. LAYER	WALL TEMPERATURE TWI	ENTHALPY REJECTION PER UNIT TUBE LENGTH BTU/(HR FT)	FRACTION OF TOTAL
.000	1.00000	1.0000	1.0000	.9622	.9622	77.247403	.00000
.250	.99961	.9857	.9674	.8994	.8994	138.934032	.25126
.500	.99923	.9732	.9457	.8822	.8822	129.805758	.54745
.750	.99886	.9620	.9162	.8600	.8600	114.752011	.80597
1.000	.99848	.9558	.8988	.8720	.8720	54.731251	1.00000
(EXIT)							

STEADY STATE IS REACHED

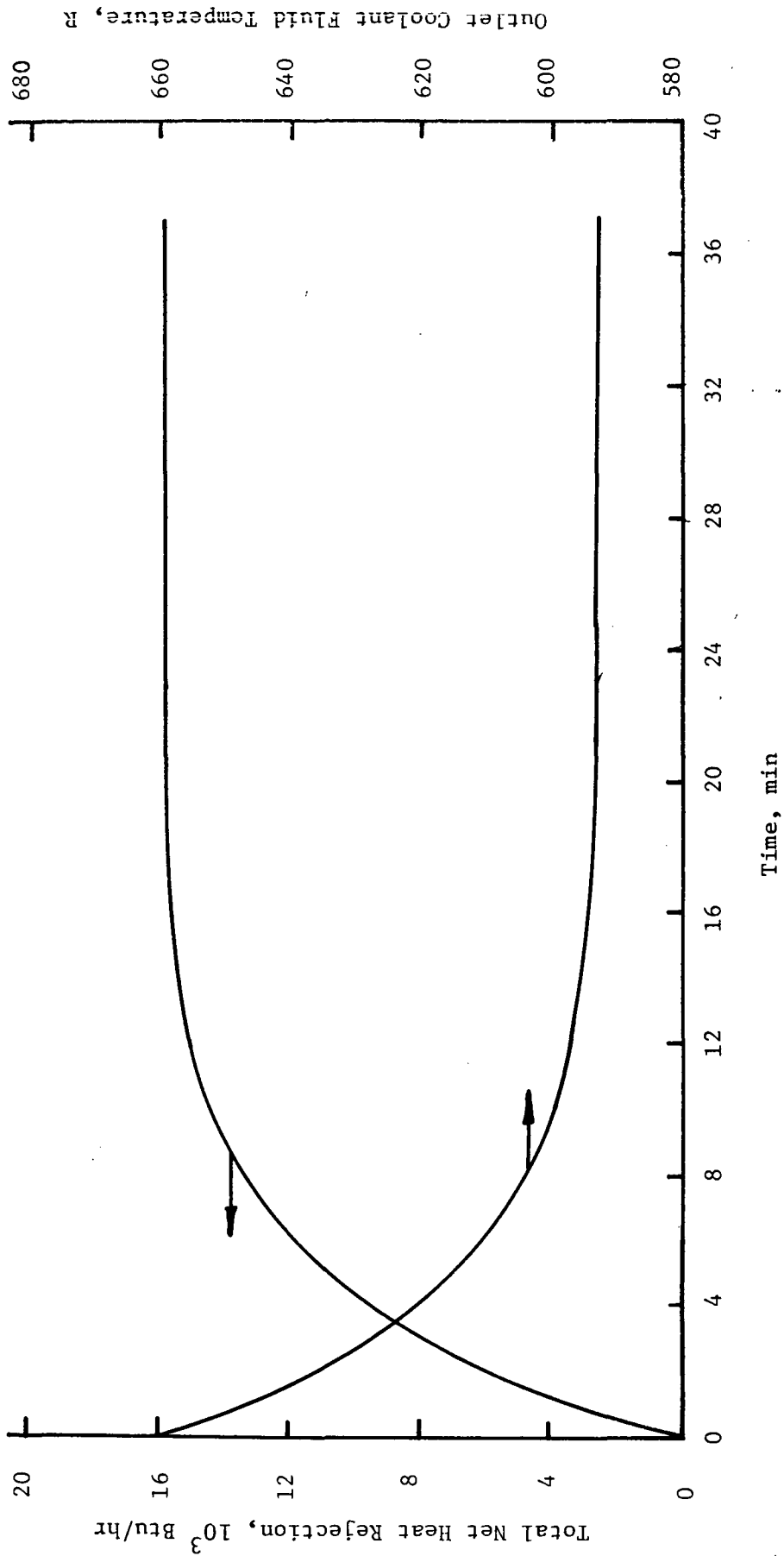


Fig. 17(b) System Response for Steady State Environment-System Heat Rejection And Outlet Fluid Temperature

## B. Program Description

### 1. Structure

The program structure is dictated by the calling sequence of the Runge-Kutta-Simpson integration procedure (see Section III-2 of the System Analysis Manual). The Source Deck consists of two parts. The first part is referred to as the Permanent Part and the second part is called the Selective Part. The Input Data Set follows the Selective Part.

#### a. The Permanent Part

The Permanent Part requires no attention from the user unless he plans major program modifications such as consideration of geometrical configurations other than the one considered on the present analysis. The Permanent Part contains the first 42 program units.

#### b. The Selective Part

The Selective part contains the options for selection of system materials. It is assembled from case to case by the user in accordance with his choice of the coolant fluid material, flow channel material, meteoroid protection layer material, fin panel material and thermal control coating material. There are always 20 program units in the Selective Part.

#### c. The Input Data Set

The Input Data Set follows immediately after the Source Deck. As discussed in Section A-3 the Data Set consists of six groups that specify the non-symmetrical panel conditions, ascent and reentry profiles, the irradiation on the panel, inlet coolant fluid conditions, system specifications and finally the program options. These separate data units are arranged in an order of increasing possibility of input changes so that data units most likely to be changed are at the end of the Data Set.

## 2. Fastrand Execution

A run procedure was written for the purpose of entering the entire program, including property subprograms, on FASTRAND drums and collecting, or mapping, the program into convenient elements. The elements are chosen so that they form commonly used groups of subprograms which then may be selectively collected into a complete program. The advantages of this technique are the significant reduction in the compilation and collection time and the elimination of submitting the program in card form.

The entire program is mapped into two basic elements: the permanent element and the property elements. The permanent element (called PERM) consists of all subroutines listed in the permanent part of the program; the property elements consist of the material properties of the surface coating, fin material, tube material, meteoroid protection material, and coolant fluid.

Each property element has a unique name. The first two letters of the name are a prefix which indicates the function of the particular material. For example, a property element starting with the prefix CF indicates that those properties to be used for the coolant fluid. The remaining letters of the element name are a suffix which describes the particular material to be used. For example, the suffix HE indicates the material helium. The prefix-suffix combination CFHE, therefore, denotes the coolant fluid of helium.

The sixteen names designating the permanent part of the program and the property elements are as follows:

### i) Permanent Elements

1. Element PERM contains all subroutines in the permanent part of the program. (See Part B.3 a through g)

### ii) Property Elements

#### A. Coolant Fluid Properties (Prefix CF)

2. Element CFHE contains all helium properties
3. Element CFSIL contains all silicon oil properties
4. Element CFNAK contains all NaK properties

5. Element CFFC43 contains all properties of the 3-M fluoro-chemical liquid FC43
  6. Element CFFC75 contains all properties of the 3-M fluoro-chemical liquid FC75
- B. Fin Properties (Prefix FN)
7. Element FNAL contains all aluminum properties
  8. Element FNCU contains all copper properties
  9. Element FNBR contains all beryllium properties
- C. Tube Properties (Prefix TB)
10. Element TBAL contains all aluminum properties
  11. Element TBCU contains all copper properties
  12. Element TBBR contains all beryllium properties
- D. Meteoroid Protection Properties (Prefix MP)
13. Element MPAL contains all aluminum properties
  14. Element MPCU contains all copper properties
  15. Element MPBR contains all beryllium properties
- E. Surface Coating Properties (Prefix SC)
16. Element SCZ93 contains all zinc oxide coating (Z93) properties

a. Control Statements

The run deck consists of the following control statements:

The catalogue card

@ CAT,P NASA.,F2

is used to create a file, called here NASA. This card is to be used only the first time the FASTRAND program is run. It must be omitted when subsequent runs are made because the file NASA should be created only once.

The assign card

@ ASG,AX NASA.

is used to name an external file which has been previously catalogued and to cause its assignment to the run. The option A specifies that



the file is currently catalogued and the option X specifies that the run is to have exclusive use of the file until the run terminates so that no other run may interfere with the program operation.

The use card

@ USE N.,NASA.

allows the user to refer to the previously catalogued file called NASA by the single letter N. The abbreviated file name will save a great deal of card punching for the subsequent portions of the run.

The Fortran compiler cards, one of which would be

@ FØR,IS N.MAIN

creates Fortran symbolic and relocatable elements both called MAIN which are to be entered into the file NASA. The I option causes the insertion of the element MAIN and the S option produces a listing of the element. There must be a FØR card for each subprogram that is to be entered into the file NASA.

After all of the subprograms have been filed, they are mapped into separate elements. This is achieved by the MAP and IN card.

The MAP cards, one of which would be

@ MAP,ISR ,N.PERM

specifies that the collector should combine a set of relocatable elements from the file NASA into a single relocatable element called PERM. There must be a MAP card for each program element that the user expects to utilize during program operation. A listing of these seventeen elements is given at the beginning of this section.

Following each MAP card is a list of the separate subprograms that are to be placed in the element declared by the MAP card. The placing of each subprogram into the element is achieved by an IN card.

The IN card, one of which would be

IN N.MAIN

places the element MAIN from the file NASA in the element name declared by the previous MAP statement. There must be an IN card for each program element that is to be entered into the element declared by the previous MAP statement.

As an example, if the user wishes to collect the subprograms called AA, BB and CC which are located in the file NASA and place them in an element called ELEM, the proper cards would be:

```
@ MAP,ISR ,NASA.ELEM
  IN NASA.AA
  IN NASA.BB
  IN NASA.CC
```

It is necessary to place a define card at the end of each mapped element.

A define card, one of which could be  
DEF THCFN, DTHCFN, CPFN

is used to list those external definitions to be retained by the resulting relocatable element. The DEF statement causes the collector to construct a table defining the entry points to the element. There should be a DEF statement following each MAP card which declares the entry point to each element collected by the MAP card.

#### b. Input Data

The runstream described above enters all subprograms into the file called NASA and then subdivides the program into convenient groupings of elements. Once this run has been made, it becomes a relatively simple task to collect and run a typical system simulation. As an example, assume the user wishes output for a radiator system consisting of an aluminum fin, copper tube, beryllium meteoroid protection layer, silicon oil coolant fluid and Z93 surface coating. Assuming the user has previously declared the file NASA and he has mapped the sixteen elements discussed above, then this combination of materials could be mapped into an absolute element called VER1 by the runstream consisting of

(Run cards appropriate to the users system)

```
@ ASG,AX NASA.
@ USE N.,NASA.
@ MAP,IS ,N.VER1
  IN N.PERM
```

```

IN N.FNAL
IN N.TBCU
IN N.MPBR
IN N.CFSIL
IN N.SCZ93
END
@ XQT N.VER1

```

This deck must be followed by the data set described in Section A-3.

As a second example, if the user wishes to run the program for a system consisting of an aluminum tube, fin and protection layer with helium coolant fluid and Z93 surface coating the deck would consist of

(Run cards appropriate to the users system)

```

@ ASG,AX NASA.
@ USE N.,NASA.
@ MAP,IS ,N.VER2
IN N.PERM
IN N.FNAL
IN N.TBAL
IN N.MPAL
IN N.CFHE
IN N.SCZ93
END
@ XQT N.VER2

```

This deck must be followed by the data deck described in Section A-3. In this example the element PERM plus the property elements are mapped into an absolute element called VER2. It should be pointed out that once these absolute elements have been mapped they are available for subsequent runs without being re-mapped. In future runs, if VER2 is called upon to be executed, the element consisting of PERM plus is the property elements listed above are automatically used.

### 3. Description of Program Units

This section describes each subprogram which makes up the Permanent and Selective Part of the program.

The description of each subprogram follows the same format. The name of the subprogram is given and it is categorized as either a FUNCTION or SUBROUTINE subprogram. The argument list is also stated.

The objective of each of the subprograms is stated. All input and output variables in the argument list are defined and units are given for all dimensional input and output quantities.

The calling subprograms and those subprograms called are listed for each program unit.

The program units are subdivided into groups of subprograms each of which has a common overall function. For example, all subprograms that are used for the calculation of the aerodynamic heating of the radiator panel are grouped into a common block of programs. In all, there are 62 program units with 42 units on the permanent part and 20 units in the Selective Part. The Permanent Part contains 42 program units grouped in seven sets from (a) through (g) as follows:

a) the principal integration program set with

1. MAIN, the main program unit which

- |          |  |
|----------|--|
| provides | α) input and output  |
|          | β) unit conversion   |
|          | γ) computation of design parameters                                  |
|          | δ) preparations for principal integration,                           |
| calls    | α) RKS, DERIVM, CNTLM, for principal integration                     |
|          | β) FLSTRT to establish the initial flow field                        |
|          | γ) FUNCTION subprograms for fluid and structural material properties |
|          | δ) QINCID, SHAPEF (see group c)                                      |
|          | ε) mathematical procedure subprograms,                               |

2. SUBROUTINE DERIVM (Y,DY,TIME) which is called from RKS,

- |          |  |
|----------|--|
| receives | α) the current system state variables Y                              |
|          | β) the current time TIME,  |
| provides | α) all derivatives of the variables Y with respect to time,          |
| calls    | α) CONVEC, QRAD  |
|          | β) FLSTRT to establish the flow field                                |
|          | γ) FUNCTION subprograms for fluid and structural material properties |
|          | δ) mathematical procedure subprograms,                               |

3. SUBROUTINE CNTLM (Y,DY,DX,X,NTRY,IFVD) which

- |                |  |
|----------------|--|
| is called from | RKS,                                   |
| receives       | α) all system state variables Y        |
|                | β) their derivatives DY                |
|                | γ) current time X, step size DX,       |
| provides       | α) output during integration           |
|                | β) integration step size control       |
|                | γ) integration termination criteria,   |
| calls          | α) HFL, fluid enthalpy subprogram      |
|                | β) mathematical procedure subprograms, |

b) the secondary integration program set, necessary to establish the dynamic fluid flow field, with

4. SUBROUTINE FLSTRT (RE,PR,DELTA) which

is called from           MAIN,DERIVM,  
 receives                α) RE, the mean Reynolds number  
                           β) PR, the mean Prandtl number  
                           γ) DELTA, the diameter-to-length ratio for  
                               the tube,  
 provides                α) fluid flow (initial and current) conditions  
                           β) table heading for initial conditions,  
 calls                    α) TRNSPT to compute friction factor and  
                               convective film coefficient (in non-  
                               dimensional form)  
                           β) RKSF, DERIVL, CNTLN  
                           γ) fluid flow property subprograms  
                           δ) mathematical procedure subprograms,

5. SUBROUTINE DERIVL (Y,DY,X) which

is called from           RKSF,  
 receives                α) local fluid flow variables Y  
                           β) position along flow channel axis, X,  
 provides                α) spatial derivatives DY of Y,

6. SUBROUTINE CNTLN (Y,DY,DX,X,NTRY,IFVD) which

is called from           RKSF,  
 receives                α) the flow variables Y  
                           β) their spatial derivatives DY  
                           γ) position X, interval DX along channel axis,  
 provides                α) output when initial conditions are  
                               established  
                           β) termination criteria for RKSF integration,

7. SUBROUTINE TRNSPT (RE,PRL,DELTA,FR,FNU) which computes non-dimensional friction factor and convective film coefficient, Eqs. 3.7 through 9 and Eqs. 3.16 through 18, and

is called from           FLSTRT,  
 receives                α) Reynolds number RE

- provides
  - β) Prandtl number PRL
  - γ) diameter-to-length ratio DELTA,
  - α)  $FR = 4f/\delta$ , Eq. 3.32
  - β)  $FNU = 4 N_{Nu} / (\delta N_{Re} N_{Pr})$ , Eq. 3.37,
- c) The program set used to calculate the incident net radiant flux, with,
8. SUBROUTINE QINCID which
- is called from MAIN,
- receives
  - α) MSTØTR control option for simulation of steady state (= 1) or transient (= 2) conditions
  - β) NSRD the number of radiator sides that are exposed to the environment
  - γ) PHN the angle between the reference used in the MRI file data and the normal to the fin panel,
- provides
  - α) data transfer from the MRI Incident Radiation Computer program or card input
  - β) averaging of radiant flux over circumference of tube cross-section,
9. SUBROUTINE SHAPEF which
- is called from MAIN,
- provides exchange function SS as defined by Eq. 6.15, see Appendix D,
10. SUBROUTINE TRMATX which
- is called from QRAD,
- provides the evaluation of the transfer matrix  $M_{ij}$ , defined by Eq. 6.7,
11. SUBROUTINE EXITAV which
- is called from QRAD,
- provides the evaluation of the excitation vector  $P_j$  in Eq. 6.18,
12. SUBROUTINE ABSØRB which
- is called from QRAD,
- receives
  - α) total hemispherical emittance  $\epsilon$ , see Eq. 6.5
  - β) temperature distributions on the fin and tube surfaces
  - γ) exchange function SS

- provides  $\delta$ ) functions  $x_{ij}$  and  $x_{ijk}$  defined by Eqs. 6.10 and 6.11,
- calls adsorptance matrix  $\alpha_{ij}$  as defined by Eq. 6.9,
- calls  $\alpha$ ) SHAPEF for SS
- $\beta$ ) EMIT for (T)
- $\gamma$ ) AVGEMT for  $x_{ij}$  and  $x_{ijk}$
- $\delta$ ) DEFINT for integration,
13. SUBROUTINE QRAD which
- is called from DERIVM,
- receives  $\alpha$ ) transfer matrix and excitation vector  $M_{ij}$  and  $P_j$ , respectively,
- provides  $\alpha$ ) inversion of transfer matrix
- $\beta$ ) solution of radiosity equations
- $\gamma$ ) grid mapping,
- calls  $\alpha$ ) TRMATX
- $\beta$ ) EXITAV
- $\gamma$ ) YINT, INTERP,

d) the program set used to calculate the aerodynamic heating during ascent and reentry, consisting of

14. FUNCTION ALTVEL (TIME, IOPTN, AVA, AVR, TA, TR, NA, NR) which interpolates user-supplied altitude and velocity profiles for a given instant, and
- is called from CONVEC,
- receives  $\alpha$ ) current real time, TIME, in seconds, measured from start of transition phase, i.e., ascent: TIME = 0, ALTVEL = AVA (1)  
reentry: TIME = 0, ALTVEL = AVR (1)
- $\beta$ ) ordered pairs of time and altitude (TA,AVA) and (TR,AVR) for ascent and reentry, respectively
- $\gamma$ ) NA and NR, the number of ordered pairs and elements in TA,AVA and TR,AVR, respectively



- provides data interpolation,  
 calls YINT for interpolation,
- 8) IØPTN = 0 for ascent  
       = 1 for reentry,
15. SUBROUTINE ATMØS (ELEV, TATM, CATM) which computes the atmospheric temperature and the speed of sound as functions of altitude and
- is called from CØNVEC,  
 receives  $\alpha$ ) current shuttle elevation, ELEV, in feet  
 provides atmospheric temperature, TATM in degrees R and speed of sound, CATM, in ft/sec,  
 calls YINT for interpolation, PØLY for evaluation of polynomials,
16. SUBROUTINE REFP (ELEV, REFTP, REFPR, REFVIS, REFRHØ, REFK, REFCP, REFGAM) which computes the atmospheric air properties as a function of elevation and of high speed reference temperature and
- is called from CØNVEC,  
 receives  $\alpha$ ) current orbiter elevation, ELEV, in feet  
            $\beta$ ) current high speed reference temperature REFTP, in degrees R,  
 provides atmospheric Prandtl number REFPR (dimensionless), dynamic viscosity REFVIS in lbm/(ft sec), density REFRHØ in lbm/ft<sup>3</sup>, thermal conductivity REFK in Btu/(sec ft R), specific heat at constant pressure REFCP in Btu/(lbm/R) and the ratio of specific heats REFGAM (dimensionless),  
 calls  $\alpha$ ) YINT for interpolation  
        $\beta$ ) PØLY for polynomial expansion,
17. SUBROUTINE NUS (MACHNØ, TATM, CATM, TIN, PRATM, VISATM, RHØATM, STAGX, VERTX, NSRAD, NUS1) which provides a single Nusselt number for the calculation of the aerodynamic heating of the orbiter radiator system at the current time and
- is called from CØNVEC,  
 receives  $\alpha$ ) current orbiter Mach number MACHNØ (dimensionless)  
            $\beta$ ) current atmospheric temperature TATM, in degrees R

- γ) current atmospheric velocity of sound, CATM, in ft/sec
- ε) temperature of the coolant fluid at the inlet plane TIN, in degrees R
- ζ) atmospheric Prandtl number PRATM, evaluated at the high speed reference temperature (dimensionless)
- η) atmospheric density RHØATM, evaluated at the high speed reference temperature in lbm/ft<sup>3</sup>
- θ) distance from the stagnation point on the shuttle vehicle to the midpoint of the radiator panel measured along a streamline, STAGX, in feet
- ι) overall dimension of the radiator panel measure in a direction parallel to the acceleration of gravity, VERTX, in feet
- κ) integer value indicating number of non-adiabatic sides of the radiator panel, NSRAD, i.e., NSRAD = 1 for single non-adiabatic surface or NSRAD = 2 for both sides of radiator being non-adiabatic,

provides

mean Nusselt number NUS1, for meteoroid layer and fin at current orbiter elevation during either ascent or reentry (dimensionless),

calls

YINT for interpolation,

18. SUBROUTINE CØNVEC (TIME) which computes normalized aerodynamic heating of fin and meteoroid protection layer for both ascent and reentry phases of orbiter operation and

is called from

DERIVM,

receives

- α) current real time TIME, in seconds, measured from start of transition phase; i.e., TIME must be measured with respect to same reference point as user supplied time arrays in FUNCTION ALTSH and VELSH,

provides

normalized convective flux from fin surface CØNFN (I,J), and from meteoroid protection surface CØNMP (I),

calls

- α) ALTVEL for altitude and velocity of orbiter
- β) ATMØS for atmospheric properties
- γ) REFP for atmospheric properties evaluated at high speed reference temperature

- δ) YINT for interpolation
- ε) NUS for Nusselt number,

19. FUNCTION CPAIR (T) which computes the specific heat at constant pressure for atmospheric air and

is called from           ATMØS, REFTP and CØNVEC,  
 receives                absolute temperature T in R,  
 provides                specific heat at constant pressure in  
                           Btu/lbmR,  
 calls                    PØLY for polynomial expansion,

20. FUNCTION ENTAIR (T) which computes the enthalpy of atmospheric air and

is called from           CØNVEC,  
 receives                absolute temperature T in R,  
 provides                enthalpy in Btu/lbm,

21. FUNCTION TNH (ENT) which computes the temperature of atmospheric air for a given enthalpy and

is called from           CØNVEC,  
 receives                enthalpy ENT in Btu/lbm,  
 provides                absolute temperature in R,  
 calls                    α) CPAIR for the specific heat at constant  
                           pressure of atmospheric air  
                           β) ENTAIR for the enthalpy of air,

e) the program to compute the thickness of the meteoroid protection layer

22. FUNCTION TK (GAMMA, A, BETA, DENSM, THETA, PHI, AN ALPHA, VELM, PØ, TAU, DENST, W, TNN, AMAN, TIN, RØUT) which computes the thickness of the meteoroid protection layer for given environmental conditions, tube and manifold areas, experimental constants and protection layer properties and

is called from           MAIN,  
 receives                α) nondimensional experimental constant,  
                           GAMMA, see Eq. 8.9

- β) nondimensional experimental constant A, see Eq. 8.9
  - γ) nondimensional constant which relates meteoroid flux to mass, BETA
  - δ) density of the meteoroid particle, DENSM, in gm/cm<sup>3</sup>
  - ε) nondimensional experimental constant THETA, see Eq. 8.9
  - ζ) nondimensional experimental constant PHI, see Eq. 8.9
  - η) nondimensional constant AN, used to describe penetration depth as a function of angle of incidence of meteoroid particle
  - θ) velocity of meteoroid particle relative to radiator panel, VELM, in ft/sec
  - ι) probability of no damage caused by impact of meteoroid, PØ
  - κ) time TAU, the radiator panel is exposed to meteoroid environment, in days
  - λ) density of protection material DENST, in lbm/ft<sup>3</sup>
  - μ) axial length of single flow channel exposed to meteoroid environment, W, in inches
  - ν) integer number of flow channels, TNN,
  - ξ) area of the manifold that is exposed, AMAN, in ft<sup>2</sup>
  - φ) temperature of coolant fluid at inlet plane of flow channel, TIN, in degrees R
  - π) outside radius RØUT of the unprotected flow channel, in inches,
- provides meteoroid protection thickness TK, in inches,
- calls ELAS for modulus of elasticity of protection material,

f) the program set which calculates the location of the adiabatic plane between two flow channels under conditions of non-symmetrical loading of the tubes and for curved radiator panels, consisting of

23. SUBROUTINE SHADE which zeros those elements in the array Q which correspond to portions of the radiator panel that do not receive solar, albedo and planetary irradiation and

is called from TCALC and QINCID,  
 receives  $\alpha$ ) NS, number of flat sides to the polygon  
 used as a reference body by MRI program  
 for calculating incident radiant flux  
 $\beta$ ) radiant flux array Q in Btu/hr ft<sup>2</sup>.  
 Array must have NS or greater values,  
 provides zeros in part of the Q array which correspond  
 to fin elements in the shade,

24. SUBROUTINE TCALC which calculates the equivalent sink temperatures  
 from incident flux data and,

is called from MAIN,  
 receives NTF the number of flat fin segments.

provides TSTAR the equivalent sink temperature in R,  
 calls SHADE for incident flux data,

25. SUBROUTINE EFFICY which evaluates the one-dimensional fin  
 efficiency and its derivative as a function of the fin base  
 temperature and

is called from TTIPS,  
 receives  $\alpha$ ) the dimensionless conductance parameter NC  
 $\beta$ ) the derivative of the conductance parameter  
 with respect to the base temperature NCPR  
 in R<sup>-1</sup>  
 $\gamma$ ) the ratio of the sink to base temperature  
 TS $\phi$ TB  
 $\delta$ ) the fin base temperature TB in R,  
 provides  $\alpha$ ) the fin efficiency ETA  
 $\beta$ ) the derivative of the fin efficiency with  
 respect to the base temperature ETAPR in  
 R<sup>-1</sup>,  
 calls P $\phi$ LY for polynomial expansion,

26. SUBROUTINE TTIP which computes the tip temperature of a one-  
 dimensional fin and

is called from ADIABH,  
 receives  $\alpha$ ) the dimensionless conductance parameter NC

- $\beta$ ) the ratio of the sink to base temperature  $TS\theta TB$ ,  
 provides the ratio of the difference between the fin tip temperature and sink temperature to the difference between the base and sink temperature,  
 calls  $P\theta LY$  for polynomial expansion,
27. FUNCTION NUSA which evaluates the Nusselt number for the coolant fluid flow for both laminar and turbulent flow and  
 is called from  $ADIABH$  and  $TTIPS$ ,  
 receives  $\alpha$ ) the Reynold number for the fluid  $REY$   
 $\beta$ ) the Prandtl number for the fluid  $PR$   
 $\gamma$ ) the ratio of the internal tube diameter to tube length  $D\theta L$ ,  
 provides the fluid Nusselt number,
28. SUBROUTINE TTIPS which uses the Newton-Raphson method to numerically solve for the base temperature of the one-dimensional fin and  
 is called from  $ADIABH$ ,  
 receives  $\alpha$ ) the integer  $IT$  used to specify the tube number for which the fin tip temperatures are calculated. Numbering system is consistent with the input parameters  $TIN$  (inlet temperature),  $PIN$  (inlet pressure),  $T$  (fin thickness),  $AL$  (tube length) and  $M$  (mass flow rate)  
 $\beta$ ) the integer  $ITST$  which designates the sink temperature data stored in the array  $TSTAR$   
 $\gamma$ ) the distance  $H1$  to the adiabatic plane for the fin attached to the left of the tube number  $IT$ , in inches  
 $\delta$ ) the distance  $H2$  to the adiabatic plane for the fin attached to the right of the tube number  $IT$ , in inches,  
 provides  $\alpha$ ) the adiabatic plane temperature for the fin attached to the right of tube  $IT$  in  $R$   
 $\beta$ ) the adiabatic plane temperature for the fin attached to the left of tube  $IT$  in  $R$ ,  
 calls  $\alpha$ )  $NUS$  for the coolant fluid Nusselt number  
 $\beta$ )  $EFFICY$  for fin efficiency  
 $\gamma$ )  $TTIP$  for the fin tip temperature,

29. SUBROUTINE ADIABH which determines the position of the adiabatic planes so that there is a continuity in adiabatic plane temperature of the one-dimensional fin and so that the heat convected from the fluid equals the heat radiated from the fin and
- is called from           MAIN,
- receives                   the integer ITST which designates sink temperature data stored in the array TSTAR,
- provides                   printout of table headings and system parameters when radiator is non-symmetrically loaded and diagnostic printout if distance to adiabatic plane exceeds the spacing between the tubes. For this case one tube is transferring heat to the fin while the adjacent tube is gaining heat. TTIPS also provides the distances to all the adiabatic planes within the non-symmetrical radiator section,
- calls                      α) property subroutines for fin and fluid properties
- β) NUS for the coolant fluid Nusselt number
- γ) TTIPS for the tip temperature of the fin
- δ) PDERIV for determination of the elements of the matrix  $P_{ij}$
- ε) FMINV for matrix inversion,
30. FUNCTION PDERIV which calculates the partial derivative  $\partial\delta_i/\partial h_j$  (see Section II-14 of the System Analysis Manual) and
- is called from           ADIABH,
- receives                   α) the integer J denotes the tube under consideration, J = -1 denotes tube IT - 1; J = 0 denotes tube IT and J = +1 denotes tube IT + 1
- β) the integer IT which specifies the tube number to be considered
- γ) the integer ITST which designates sink temperature data stored in the array TSTAR,
- provides                   the partial derivative  $\partial h_i/\partial h_j$
- calls                      TTIPS for evaluation of the temperature at the adiabatic plane,

- g) the program set consisting of all the mathematical procedures discussed in detail in Part III of the System Analysis Manual
31. SUBROUTINE RKSF (see Section III.15) which
    - is called from FLSTRT, for dynamic fluid flow field,
    - calls DERIVL, CNTLN,
  32. SUBROUTINE RKS (see Section III.15) which
    - is called from MAIN, for principal integration,
    - calls DERIVM, CNTLM,
  33. FUNCTION POLY (X,A,M) (see Section III.16),
  34. FUNCTION YINT (X,Y,M,N,P) (see Section III.17),
  35. SUBROUTINE DDX (Y,DY,DX,N), (see Section III.18)
  36. SUBROUTINE D2DX2 (Y,D2Y,DX,N),
  37. FUNCTION DEFINT (Y,DX,N), (see Section III.19)
  38. SUBROUTINE FINT (Y,YO,DX,N,F),
  39. SUBROUTINE INTERP (NX1,MZ1,XX1,ZZ1,YY1,NX2,MZ2,XX2,ZZ2,YY2) maps a two-dimensional function YY1 from one grid (XX1,ZZ1) onto another grid (XX2,ZZ2) and
    - is called from QRAD,
    - receives
      - α) the number of nodal points (NX1,MZ1) of the original grid
      - β) the function YY1
      - γ) the number of nodal points (NX2,MZ2) of new grid,
    - provides the function YY2 at the new grid,
    - calls YINT,
    - is restricted to  $NX1,MZ1,NX2,MZ2 \leq 10$ ,
  40. FUNCTION DEFNT serves to integrate by the trapezoidal rule and is used where random error accumulation is more critical than truncation errors, and V accepts through its argument list (Y, DX, N)



- $\alpha$ ) the array Y of N equally spaced ordinates
  - $\beta$ ) the interval DX
  - $\gamma$ ) the integer constant N, representing the number of nodal points.
- returns the integral of YCS) as DEFNT.

41. SUBROUTINE MTXINV (N,M) performs two related tasks:

- (i) to solve a system of N linear algebraic equations, when  $M = N + 1$ ,
- (ii) to invert an N x N invertible matrix, when  $M = 2N$ .

For task (i) it

accepts through COMMON TRMTX (k,1) the augmented coefficient matrix (see Chapter III.7). Here k and 1 represent integer constants,  $k = N$ ,  $1 = N + 1$ .

through the argument list (N,M) the rank N of the coefficient matrix and  $M = N + 1$ , two integer constants.

returns the solution in TRMTX (I,J),  $I = 1, \dots, k$ ,  $J = k$ .

For task (ii) it

accepts through COMMON TRMTX (k,1)

the  $k \times k$  matrix in the first k columns of TRMTX (I,J),  $I = 1, \dots, k$ ;  $J = 1, \dots, k$  and the  $k \times k$  identity matrix in the second k columns of TRMTX (I,J)  $I = 1, \dots, k$ ;  $J = k + 1, \dots, 2k$ .

Hence  $1 = 2k$ .

through the argument list (N,M) the rank N of the coefficient matrix and  $M = 2N$ , both integer constants.

returns the inverted matrix in the second k columns of TRMTX (I,J),  $I = 1, \dots, k$ ;  $J = k + 1, \dots, 2k$ .

42. SUBROUTINE FMINV serves to solve N linear algebraic equations by the method used for SUBROUTINE MTXINV. Its calling provisions are designed to accommodate the requirements of ADIABH. It

receives, through its argument list (A,X,N,M)

$\alpha$ ) the (N x N), invertible coefficient matrix A

$\beta$ ) the known, N-dimensional vector X

$\gamma$ ) the rank (N) of A, an integer constant

$\delta$ )  $M = N + 1$

where  $N \leq 25$

returns the solution  $Y = A^{-1}X$  placed in the array X

is called from ADIABH.

The Selective Part contains five groups of thermophysical properties, with the total of 20 program units. Each group constitutes a property package, one each for

- (i) the coolant fluid (8 program units)
- (ii) the flow channel material (3 program units)
- (iii) the meteoroid protection layer (4 program units)
- (iv) the fin material (3 program units)
- (v) the thermal control coating (2 program units).

The choice of a particular material combination must be reflected in the corresponding data specification of the Input Data Set, (see Section 3f of Chapter A).

- a) The coolant property subprogram set consists of eight elements, 43 through 50
- 43. FUNCTION RHØF (P,T) which computes the fluid density RHØF in slug/ft<sup>3</sup> as a function of pressure P in lbf/ft<sup>2</sup>, and absolute temperature T in R,
  - 44. FUNCTION BETA (RHØ,T) which computes the isobaric expansion coefficient  $\beta$ , Section II.C.11, in 1/R as a function of density RHØ in slug/ft<sup>3</sup>, and temperature T in R,
  - 45. FUNCTION CAPPA (P,T) which computes the isothermal compressibility, Eq. 11.3, in ft<sup>2</sup>/lbf, as a function of pressure P in lbf/ft<sup>2</sup>, and temperature T in R,
  - 46. FUNCTION CPF (RHØ,T) which computes the specific heat at constant pressure CPF, in Btu/(slug R), as a function of density RHØ in slug/ft<sup>3</sup>, and temperature T, in R,
  - 47. FUNCTION HFL (RHØ,T) which computes the fluid enthalpy HFL, in Btu/slug, as a function of density RHØ in slug/ft<sup>3</sup>, and temperature T, in R,
  - 48. FUNCTION VISC (RHØ,T) which computes the fluid dynamic viscosity VISC, in slug/(ft sec) as a function of density RHØ in slug/ft<sup>3</sup>, and temperature T, in R,

49. FUNCTION THCF (RHØ,T) which computes the fluid thermal conductivity THCF in Btu/(hr ft R) as a function of density RHØ in slug/ft<sup>3</sup>, and temperature T, in R,
50. FUNCTION PF (RHØ,T) which computes the fluid pressure PF in lbf/ft<sup>2</sup> as a function of density RHØ in slugs/ft<sup>3</sup>, and absolute temperature T, in R. For specific details of the coolant fluid properties, see Section II.11 and 12 and Appendix B of the System Analysis Manual.
- b) The coolant channel material property subprogram set consists of three elements, 51 through 53.
51. FUNCTION THCTB (T) computes the thermal conductivity THCTB in Btu/(hr ft R), of the tube wall material as a function of the absolute temperature T, in R,
52. FUNCTION DTHCTB (T) computes the relative change of thermal conductivity k with temperature T, namely  $(dk_w/dT)/k_w$  in 1/R, as a function of temperature T, in R,
53. FUNCTION CPTB (T) computes the specific heat at constant pressure CPTB, in Btu/(slug R) as a function of T, in R.

For specific details of the coolant channel properties, see Appendix A of the System Analysis Manual.

- c) the meteoroid protection layer material properties are coded in a set of four elements, 54 through 57
54. FUNCTION THCMP (T),
55. FUNCTION DTHCMP (T),
56. FUNCTION CPMP (T),
57. FUNCTION ELAS (T) computes the modulus of elasticity ELAS in lbf/in<sup>2</sup> as a function of temperature T, in R.

For specific details of the meteoroid protection properties, see Appendix A of the System Analysis Manual.

- d) The fin material properties required are coded in three FUNCTION subprograms, numbered 58 through 60

58. FUNCTION THCFN (T) computes the thermal conductivity THCFN, in Btu/(hr ft R) for the fin material as a function of absolute temperature T, in R,
59. FUNCTION DTHCFN (T) computes the relative change of thermal conductivity, namely  $(dk_f/dT)/k_f$  in 1/R, as a function of temperature T, in R,
60. FUNCTION CPFN (T) computes the specific heat at constant pressure CPFN, in Btu/(slug R) as a function of temperature T, in R.

For specific details of the fin material properties, see Appendix A of the System Analysis Manual.

- e) Optical properties of the thermal control coating are coded in two subprograms, numbered 61 and 62

61. FUNCTION EMIT (T) computes the total hemispherical emittance in accordance with Eq. 6.5 of the System Analysis Manual as a function of the surface temperature T in R, and is called by ABSORB and EXITAV,
62. SUBROUTINE AVGEMT (T,XX,XXX,N) evaluates the expressions for  $x_{ij}$  and  $x_{ijk}$  given as Eqs. 6.10 and 11 in the System Analysis Manual for N interacting surface elements. It is called by ABSORB (see program No. 12).

For specific details of surface coating properties, see Appendix C of the System Analysis Manual.

This completes the Source Deck discussion. Recall that the Source Deck composition implies a particular selection of coolant fluid and structural materials.

#### 4. Program Listing

This section contains the entire program listing in alphabetical order of subprogram name.

Labelled common blocks and external program references and printed out prior to the listing of each program unit.

All program units have been documented with comment cards for the convenience of the user.

All program units are summarized in a table of contents which follows the final unit listing. The table of contents lists all program elements as either symbolic or relocatable and it also prints the time and date when the program element was most recently stored on FASTRAND drum.

SUBROUTINE ABSORB ENTRY POINT 000552

STORAGE USED: CODE(1) 000600; DATA(0) 000115; BLANK COMMON(2) 000000

COMMON BLOCKS:

0003 GRD 003721  
0004 ABSRST 000601  
0005 AVGABS 000251

EXTERNAL REFERENCES (BLOCK, NAME)

0006 EMIT  
0007 DEFNT  
0010 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000032	116G	0001	000042	123G	0001	000062	132G	0001	000065	135G	0001	000067	140G
0001	000137	152G	0001	000155	161G	0001	000156	164G	0001	000244	174G	0001	000251	177G
0001	000260	202G	0001	000271	206G	0001	000352	220G	0001	000355	223G	0000	000357	226G
0001	000430	240G	0004	R 000000	ALPHFN	0004	R 000226	ALPHMP	0000	R 000000	AUX1	0000	R 000005	AUX2
0000	R 000012	AUX3	0000	R 000017	AUX4	0003	R 000365	CSF	0007	R 000000	DEFNT	0003	R 0003464	DX2
0003	003472	DX21	0003	003463	DZMFN	0003	R 003410	EBFN	0003	R 003441	ERMP	0006	R 000000	EMIT
0004	R 000461	EMITFN	0004	R 000512	EMITP	0003	R 003467	EXTIFN	0003	R 003471	EXTIMP	0003	R 003466	EXTSFN
0003	R 003470	EXTS4P	0004	000517	EXTVCT	0000	I 000032	I	0000	I 000037	IL	0000	000057	INJPS
0000	I 000034	J	0000	I 000040	JL	0000	I 000036	K	0000	I 000043	L	0003	003461	LCT
0003	003462	LTT	0003	003460	MCVRD	0003	R 003473	SS	0003	R 003670	SSTT	0005	R 000100	TFIN
0000	R 000035	TFIN1	0000	R 000033	TM	0005	R 000244	TPR	0003	R 000000	TRMTX	0000	R 000041	X
0005	R 000062	XXFN	0005	R 000000	XXMP	0005	R 000071	XXFN	0005	R 000031	XXAMP	0003	003446	XX2
0000	R 000042	Y	0000	R 000026	ZCF	0000	R 000031	ZCM	0000	R 000024	ZF	0000	R 000025	ZFM
0000	R 000027	ZM	0000	R 000030	ZWM	0003	003453	ZZ2						

00101 1\*  
00101 2\*  
00101 3\*  
00101 4\*  
00101 5\*  
00101 6\*  
00101 7\*  
00103 8\*  
00103 9\*  
00103 10\*  
00104 11\*  
00104 12\*  
00105 13\*  
00105 14\*

C  
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C

SUBROUTINE ABSORB (T0, XCFN, XCMP, XCFN, XXCMP)

THIS SUBROUTINE COMPUTES:  
THE ABSORBANCE MATRIX REPRESENTING THE MAIN ABSORBANCE AFTER  
INTERNAL INTEGRATIONS

COMMON /GRD/ TRMTX(30,60),EBFN(5,5),EBMP(5),XX2(5),ZZ2(5),  
1 MCVRD,LCT,LTT,DZMFN,  
2 DX2,CSF,EXTSFN,EXTIFN,EXTSMP,EXTIMP,DX21,SS(5,5,5),  
3 ,SSTT(5,5)  
COMMON /ABRST/ALPHFN(5,5,6),ALPHMP(5,31),EMITFN(5,5),EMITP(5),  
1 EXTVCT(50)  
COMMON/AVGABS/XXMP(5,5),XXAMP(5,5),XXFN(7),XXXFN(7),  
1 TFIN(10,10),TPR(5)

\*\*\*\*\* ABSORB \*\*\*\*\*

```

00106 DIMENSION AUX1(5),AUX2(5),AUX3(5),AUX4(5)
00107 ZF = EXTSFN*EXTIFN
00108 ZFM = ZF*(1-XCFN)
00109 ZCF = ZF*(XCFN-XXCFN)
00110 ZM = EXTSMP*EXTIMP
00111 ZMM = ZM*(1-XCMP)
00112 ZCM = ZM*(XCMP-XXCMP)
00113 FIN
00114 DO 10 I=1,5
00115 TM = 10*TPR(I)
00116 EMITMP(I) = EMIT(TM)
00117 DO 10 J=1,5
00118 TFINN = 10*TFIN(J,I)
00119 10 EMITFN(J,I) = EMIT(TFINN)
00120 DO 60 K=1,5
00121 DO 30 IL=1,5
00122 DO 20 JL=1,5
00123 X = CSF*SS(JL,IL,K)*EBFN(IL,JL)
00124 AUX1(JL) = X*(XXMP(IL,JL)-XXAMP(IL,JL))
00125 AUX3(JL) = X*(EMITFN(IL,JL)-XXMP(IL,JL))
00126 AUX2(IL) = DEFNT (AUX1,DX2,5)+AUX1(1)*DX2/2.0
00127 30 AUX4(IL) = DEFNT (AUX3,DX2,5)+AUX3(1)*DX2/2.0
00128 DO 35 IL=1,5
00129 Y = SST(IL,K)*EBMP(IL)
00130 AUX1(IL) = Y*(XFN(IL)-XXFN(IL))
00131 35 AUX3(IL) = Y*(EMITMP(IL)-XFN(IL))
00132 DO 60 I=1,5
00133 DO 60 J=1,5
00134 ALPHA(I,J,K) = (XFN(K)*EBMP(K)+DEFNT(AUX1,0.25,5)+DEFNT(AUX2,
00135 0.25,5)+ZCM)/(EMITMP(K)*EBMP(K)+DEFNT(AUX3,0.25,
00136 5)+DEFNT(AUX4,0.25,5)+ZMM)
00137 1
00138 2
00139 60 ALPHA(I,J,6) = XCFN
00140 CHANNEL
00141 DO 120 K=1,5
00142 DO 90 I=1,5
00143 DO 90 J=1,5
00144 L = J+5*(I-1)
00145 DO 80 JL=1,5
00146 X = SS(I,J,JL)*EBMP(JL)
00147 AUX1(JL) = X*(XFN(JL)-XXFN(JL))
00148 90 AUX3(JL) = X*(EMITMP(JL)-XFN(JL))
00149 C
00150 90 ALPHA(K,L) = (XMP(J,I)*EBFN(J,I)+DEFNT(AUX1,0.25,5)+ZCF)/
00151 (EMITFN(J,I)*EBFN(J,I)+DEFNT(AUX3,0.25,5)+ZFM)
00152 DO 95 I=1,5
00153 DO 93 JL=1,5
00154 DO 92 J=1,5
00155 X = CSF*SS(JL,IL,K)*EBFN(IL,JL)
00156 AUX1(JL) = X*(XXMP(IL,JL)-XXAMP(IL,JL))
00157 71*

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***** ABSORB *****
00232 72* 92 AUX3(JL) = X*(EMITFN(IL,JL)-XXMP(IL,JL))
00234 73*   AUX2(IL) = DEFNT (AUX1,DX2,5)+AUX1(1)*DX2/2.0
00235 74*   AUX4(IL) = DEFNT (AUX3,DX2,5)+AUX3(1)*DX2/2.0
00237 75*   DO 94 IL=1,5
00242 76*     Y = SSTT(IL,I)*EBMP(IL)
00243 77*     AUX1(IL) = Y*(XXFN(IL)-XXFN(IL))
00244 78*     AUX3(IL) = Y*(EMITMP(IL)-XXFN(IL))
00246 79*   94 AUX3(IL) = Y*(EMITMP(IL)-XXFN(IL))
00246 80*   95 ALPHMP(K,25+I) = (XXFN(I)*EBMP(I)+DEFNT(AUX1,0.25,5)+DEFNT(AUX2,
1     0.25,5)+ZCN)/(EMITMP(I)*EBMP(I)+DEFNT(AUX3,0.25,
2     5)+DEFNT(AUX4,0.25,5)+ZMN)
C
00246 81*
00246 82*
00250 83*   120 ALPHMP(K,31) = XCMP
00252 84*   RETURN..
00253 85*   END

```

END OF COMPILATION: NO DIAGNOSTICS.

QHDG:P \*\*\*\*\* ADIABH \*\*\*\*\*

\*\*\*\*\* ADIABH \*\*\*\*\*

QFOR,S ME\*NASAS5,ADIABH,ME\*NASAS5,ADIABH  
FOR S9A-07/13/72-20:52:57 (0,)

SUBROUTINE ADIABH ENTRY POINT 000506

STORAGE USED: CODE(1) 000522; DATA(0) 000451; BLANK COMMON(2) 000000

COMMON BLOCKS:

0003 ADBH 000352

EXTERNAL REFERENCES (BLOCK, NAME)

0004 NUSA  
0005 RHOF  
0006 VISC  
0007 CPF  
0010 THCF  
0011 THCFN  
0012 EMIT  
0013 TTIPS  
0014 PDERIV  
0015 FMINV  
0016 NPRT\$  
0017 NIO2\$  
0020 NSTOPS  
0021 EXP  
0022 NWDIUS  
0023 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000	000174	1005F	0001	000004	1136	0001	000074	131G	0000	000373	15F	0001	000124	150G									
0001	000161	157G	0001	000167	165G	0001	000201	173G	0001	000271	205G	0001	000326	231G									
0001	000401	255G	0001	000420	263G	0001	000445	276G	0000	000316	5F	0001	000077	55L									
0001	000115	60L	0001	000322	76L	0001	000346	81L	0001	000336	815L	0001	000356	82L									
0001	000362	86L	0001	000435	89L	0001	000441	895L	0000	R	000000	A	0003	R	000001	AL							
0003	R	000161	CONF	0003	R	000173	CONFNA	0003	R	000147	CP	0007	R	000000	D	0003	R	000000	D				
0000	R	000144	DHAB	0003	R	000326	DIL	0003	R	000244	EFF	0003	R	000205	EMIS	0012	R	000000	EMIT				
0003	R	000013	H	0003	R	000231	HAB	0000	R	000170	HABR	0000	I	000156	I	0000	I	000163	ICNT				
0000	000425	INJP\$	0000	I	000164	IT	0000	I	000173	J	0000	I	000171	J1	0000	I	000172	J2	0000	I	000161	NT1	
0003	R	000064	M	0000	I	000160	NSTOP	0003	I	000076	NT	0000	I	000162	NTM1	0000	R	000157	RHO	0010	R	000000	THCF
0004	I	000000	NUSA	0014	R	000000	PDERIV	0003	R	000052	PIN	0003	R	000314	QOUT	0000	R	000026	TH	0000	R	000165	TL1
0005	R	000000	RHOF	0003	R	000217	TB	0000	R	000167	TF	0003	R	000026	TH	0000	R	000165	TL1	0003	R	000135	VSC
0011	R	000000	THCFN	0003	R	000040	TIN	0003	R	000270	TL	0006	R	000000	VISC								
0003	R	000340	TOUT	0003	000077	TSTAR																	

00101 1\* SUBROUTINE ADIABH(ITST)  
00103 2\* PARAMETER NTP = 10  
00104 3\* PARAMETER NTP1 = NTP+1

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00105 4* PARAMETER NTPSQ = NTP*NTP
00106 5* REAL M
00107 6* COMMON /ADBH/ D,AL(NTP),H(NTP),TH(NTP),TIN(NTP),PIN(NTP),
00107 7* M(NTP),NT,TSTAR(NTP,3),VSC(NTP),CP(NTP),CONFL(NTP),
00107 8* CONFNA(NTP),EMIS(NTP),TB(NTP),HAB(NTP1),EFF(NTP,2),TL(NTP,2),
00107 9* GOUT(NTP),DTL(NTP),TOUT(NTP)
00110 10* DIMENSION A(NTPSQ),DHAB(NTP)
00111 11* DEFINE U(I) = 3.1415927*CONFL(I)*AL(I)*NUSA(48.0*M(I))/(3.1415927*
00111 12* ID*VSC(I)),VSC(I)*CP(I)/CONFL(I),D/(12*AL(I))/(M(I)*CP(I))
00112 13* DO 10 I = 1,NT
00115 14* RHO = RHOF(PIN(I),TIN(I))
00116 15* VSC(I) = VISC(RHO,TIN(I))
00117 16* CP(I) = CPF(RHO,TIN(I))
00120 17* CONFL(I) = THCF(RHO,TIN(I))
00121 18* CONFNA(I) = THCFN(TIN(I))
00122 19* EMIS(I) = EMIT(TIN(I))
00124 20* NSTOP=20
00125 21* NT1=NT+1
00126 22* NTM1=NT-1
00127 23* ICNT=1
00130 24* DO 2 IT=1,NT1
00133 25* HAB(IT)=H(IT)
00135 26* CONTINUE
00136 27* ICNT=ICNT+1
00137 28* IF(ICNT.LE.NSTOP)GO TO 60
00141 29* PRINT 1005,NSTOP
00144 30* FORMAT('PROGRAM STOPPED IN SUBROUTINE ADIABH BECAUSE THE DISTANCE
00144 31* > TO AN',/, 'ADIABATIC PLANE FAILED TO CONVERGE WITHIN THE ALLOWABL
00144 32* >E RANGE OF
00144 33* >',/, 'VALUES(2.5% AND 97.5% OF TOTAL FIN WIDTH) AFTER',I3,' ITERAT
00144 34* >IONS',/, 'THIS IS PROBABLY BECAUSE THE INPUT CONDITIONS DO NOT AL
00144 35* >LOW A SOLUTION',/, 'IN THIS RANGE. HOWEVER THE NUMBER OF ITERATION
00144 36* >S CAN BE INCREASED',/, ' BY ASSIGNING A LARGER VALUE TO NSTOP IN S
00144 37* >UBROUTINE ADIABH.',)
00145 38* STOP
00146 39* CONTINUE
00147 40* DO 6 IT=1,NT
00152 41* CALL TTIPS(IT,ITST,HAB(IT),H(IT+1)*2.0-HAB(IT+1),TL1,TL2)
00153 42* TL(IT,1) = TL1
00154 43* TL(IT,2) = TL2
00156 44* DO 7 IT=1,NTM1
00161 45* DTL(IT)=TL(IT,2)-TL(IT+1,1)
00162 46* CONTINUE
00164 47* DO 75 IT=1,NTM1
00167 48* DO 75 IT=1,NTM1
00172 49* IF(ABS(DTL(IT)).GT.0.05)GO TO 76
00175 50* DO 752 IT = 1,NT
00176 51* TF = (TIN(IT)-TB(IT))*(1.0-EXP(-U(IT)))
00177 52* TOUT(IT) = M(IT)*CP(IT)*TF
00201 53* WRITE(6,5)
00203 54* FORMAT(////,7X,'BULK TEMP',5X,'INCHES TO ADIABATIC PLANE',
00203 55* 5X,'PER CENT FIN EFFICIENCY',5X,'FIN TIP TEMP DEG R',
00203 56* 5X,'OUTLET TEMP',5X,'HEAT REJECTED',/,1X,'TUBE',4X,
00203 57* 'DEG R',11X,'LEFT',8X,'RIGHT',12X,2('LEFT',8X,'RIGHT',9X),
00204 58* 'DEG R',11X,'BTU/HR',/)
00204 59* DO 100 I = 1,NT
00207 60* HABR = H(I+1)*2.0-HAB(I+1)

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00210 61*      WRITE(15) I,IB(I),HAB(I),HABR,EFF(I),EFF(I+2),TL(I),
00210 62*      TL(I+2),TOUT(I),GOUT(I)
00225 63*      FORMAT(1X,I2,3X,1PE10.4,6X,1PE10.4,2X,1PE10.4,7X,
00225 64*      2(1PE10.4,2X,1PE10.4,4X),1PE10.4,7X,1PE10.4)
00226 65*      RETURN
00227 66*      CONTINUE
00230 67*      DO 88 IT=1,NTMI
00233 68*      IF(NT.NE.2)GO TO 815
00235 69*      J1=0
00236 70*      J2=0
00237 71*      GO TO 86
00240 72*      IF(IT.NE.1)GO TO 81
00242 73*      J1=0
00243 74*      J2=1
00244 75*      GO TO 86
00245 76*      IF(IT.NE.NTMI)GO TO 82
00247 77*      J1=-1
00250 78*      J2=0
00251 79*      GO TO 86
00252 80*      J1=-1
00253 81*      J2=1
00254 82*      DO 88 J=J1,J2
00257 83*      A(IT*NTMI-1+NTMI+J*NTMI+IT)=PDERIV(J,IT,ITST)
00262 84*      DO 8 IT=1,NTMI
00265 85*      DHAB(IT)=-DTL(IT)
00267 86*      IF(NT.NE.2)GO TO 89
00271 87*      DHAB(1)=-DTL(1)/A(1)
00272 88*      GO TO 895
00273 89*      CALL FMINV(A,DHAB,NTMI,NT)
00274 895*      CONTINUE
00275 91*      DO 9 IT=2,NT
00300 92*      HAB(IT)=HAB(IT)+DHAB(IT-1)
00301 93*      HAB(IT)=MIN(HAB(IT),1.95*H(IT))
00302 94*      HAB(IT)=MAX(.05*H(IT),HAB(IT))
00303 95*      CONTINUE
00305 96*      GO TO 55
00306 97*      END

```

END OF COMPILATION: NO DIAGNOSTICS.

SMDS6.P \*\*\*\*\* ALTVEL \*\*\*\*\*

\*\*\*\*\* ALTVEL \*\*\*\*\*

QFOR,S ME\*NASAS,ALTVEL,ME\*NASAS,ALTVEL  
FOR S9A-07/13/72-20:53:00 (0,)

FUNCTION ALTVEL ENTRY POINT 000151

STORAGE USED: CODE(1) 000207; DATA(0) 000016; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 YINT  
0004 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000052 1L 0001 000120 2L 0001 000124 3L 0001 000133 4L 0000 R 000000 ALTVEL  
0000 000002 INJ\$S 0000 I 000001 M 0003 R 000000 YINT

```

00101 1*
00101 2*
00101 3*
00101 4*
00101 5*
00101 6*
00101 7*
00101 8*
00101 9*
00103 10*
00104 11*
00106 12*
00110 13*
00111 14*
00113 15*
00115 16*
00117 17*
00120 18*
00121 19*
00122 20*
00123 21*
00124 22*
00125 23*
00126 24*

FUNCTION ALTVEL (TIME,IOPTN,AVA,AVR,TA,TR,NA,NR)
C
C THIS SUBROUTINE COMPUTES :
C ALTITUDE AND VELOCITY OF ORBITER AS A FUNCTION OF TIME FROM
C LAUNCH (IOPTN = 0 ) OR TIME FROM PASSING THROUGH 400,000 FEET
C FOR REENTRY ( IOPTN = 0 )
C
C TIME IN SECS, ALTITUDE IN FEET
C AND VELOCITY IN FEET/SEC
C
DIMENSION TA(NA),TR(NR),AVA(NA),AVR(NR)
IF (TIME.GT. TA(NA) .AND. IOPTN.EQ. 0) GO TO 3
IF (TIME.GT. TR(NR) .AND. IOPTN.EQ. 1) GO TO 4
M = 2
1 IF (IOPTN.EQ. 0) ALTVEL = YINT(TA,AVA,NA,M,TIME)
IF (IOPTN.EQ. 1) ALTVEL = YINT(TR,AVR,NR,M,TIME)
IF (ALTVEL.LT. 0.0 .AND. M.GT. 2) GO TO 2
RETURN
2 M = M-1
GO TO 1
3 ALTVEL = AVA(NA)
RETURN
4 ALTVEL = AVR(NR)
RETURN
END

```

END OF COMPILATION: NO DIAGNOSTICS.

QMDG:P \*\*\*\*\* ATMOS \*\*\*\*\*

QFOR,S ME\*NASAS.ATMOS,ME\*NASAS.ATMOS  
FOR S9A-07/13/72-20:53:03 (0,)

SUBROUTINE ATMOS ENTRY POINT 000237

STORAGE USED: CODE(1) 000261; DATA(0) 000211; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 POLY  
0004 CPAIR  
0005 NWDUS  
0006 NI02\$  
0007 SGR1  
0010 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000105	10L	0001	000137	15L	0001	000070	2L	0001	000161	20L	0001	000057	231G	
0001	000012	30L	0000	000116	35F	0001	000022	40L	0000	000136	45F	0001	000032	60L	
0000	R	000041	A	0000	R	000055	B	0000	R	000074	CF1	0000	R	000107	CPATM
0000	R	000104	DH	0000	R	000100	ELEVH	0000	R	000110	GAMATM	0000	R	000101	GEOHM
0000	R	000007	GH	0000	R	000031	GT	0000	R	000000	H	0000	I	000201	INJPS
0000	I	000105	J	0000	R	000077	MWT	0000	R	000113	PATM	0000	R	000075	PSL
0000	R	000112	PZ	0000	R	000115	RHOATM	0000	R	000076	RSI	0000	R	000073	RUI
0000	R	000114	RZ	0000	R	000020	T	0000	R	000106	TMK	0000	R	000111	Z

SUBROUTINE ATMOS (ELEV,TATM,CATM)

00101 1\*  
00101 2\*  
00101 3\*  
00101 4\*  
00101 5\*  
00101 6\*  
00101 7\*  
00101 8\*  
00101 9\*  
00101 10\*  
00101 11\*  
00101 12\*  
00101 13\*  
00101 14\*  
00101 15\*  
00103 16\*  
00104 17\*  
00104 18\*  
00104 19\*  
00104 20\*  
00104 21\*  
00104 22\*  
00104 23\*

THIS SUBROUTINE COMPUTES :  
PROPERTIES FOR ATMOSPHERIC AIR UP TO 528,000 FEET (100 MILES OR 161 KM)  
PROPERTIES FOR AN ALTITUDE LESS THAN 301,000 FEET TAKEN FROM  
U. S. STANDARD ATMOSPHERE FOR 1966. PROPERTIES FOR ALTITUDES BETWEEN  
301,000 FEET AND 528,000 FEET ARE TAKEN FROM APPROXIMATE ANALYSIS GIVEN  
IN THE SAME REFERENCE. ERRORS IN PRESSURE AND DENSITY ARE NO LARGER  
THAN 5 PERCENT FOR ALTITUDES GREATER THAN 301,000 FEET AND LESS  
THAN 528,000 FEET.

ELEVATION IN FEET  
TEMPERATURE IN DEG R.  
SPECIFIC HEAT IN BTU/LB\*M R  
VEL OF SOUND IN FT/SEC  
DIMENSION H(7),GH(9),T(9),GT(8),A(12),B(12)  
DATA H(1),H(2),H(3),H(4),H(5),H(6),H(7) / 0.0,1.0,-1.5731262E-07,2.  
14656553E-14,-3.8667054E-21,6.0621354E-28,-9.5013649E-35,GH(1),GH(  
22),GH(13),GH(4),GH(5),GH(6),GH(7),GH(8),GH(9) / 0.0,11.0,20.0,32.0,4  
37.0,52.0,61.0,79.0,90.0,T(1),T(2),T(3),T(4),T(5),T(6),T(7),T(8),T  
4(9) / 288.15,216.65,216.65,228.65,270.65,270.65,252.65,180.65,180.6  
55,GT(1),GT(2),GT(3),GT(4),GT(5),GT(6),GT(7),GT(8) / -6.5,0.0,1.0,2  
6.8,0.0,-2.0,-4.0,0.0,0.0,WMO,RUI / 28.9644,8314.32,1545.31/

```

00104 24* 7 CF1/ 1000.0/,PSL,RSL /2116.22657,0.07647438/
00154 25* DATA A(1),A(2),A(3),A(4),A(5),A(6),A(7),A(8),A(9),A(10),A(11),
00154 26* 1A(12)/
00154 27* 1 0.10E+01, 0.3533367370E-01, -0.7474788290E-03,
00154 28* 0.2121572235E-03, -0.1325255219E-04, 0.5344159692E-06,
00154 29* -0.1322745646E-07, 0.1965359762E-09, -0.1723714966E-11,
00154 30* 0.8707590786E-14, -0.2341816445E-16, 0.2597772972E-19/,
00154 31* 58(1),B(2),B(3),B(4),B(5),B(6),B(7),B(8),B(9),B(10),B(11),B(12)/
00154 32* 0.10E+01, 0.3393495800E-01, -0.3433553057E-02,
00154 33* 0.5497466428E-03, -0.3228358326E-04, 0.1106617734E-05,
00154 34* -0.2291755795E-07, 0.2902146443E-09, -0.2230070938E-11,
00154 35* 0.1010575226E-13, -0.2482089627E-16, 0.2548769715E-19/
00205 REAL MWT
00206 IF (ELEV .LT. 0.0) GO TO 30
00210 IF (ELEV .GT. 528000.) GO TO 40
00212 GO TO 60
00213 30 WRITE (6,35)
00215 35 FORMAT (1X,'AN ATTEMPT HAS BEEN MADE TO EVALUATE ATMOSPHERIC PROPE
00215 42* RTIES FOR NEGATIVE ALTITUDES')
00216 RETURN
00217 40 WRITE (6,45)
00221 45 FORMAT (1X,'AN ATTEMPT HAS BEEN MADE TO EVALUATE ATMOSPHERIC PROPE
00221 46* RTIES FOR AN ALTITUDE EXCEEDING 100 MILES')
00222 RETURN
00223 60 IF (ELEV .GT. 301000.) GO TO 20
00225 ELEV = ELEV+0.3048
00226 GEOHM = POLY(7,H,ELEV)
00227 51* GEOH = GEOHM/0.3048
00230 DO 1 I=1,9
00233 53* DH = GH(I)-GEOHM/CF1
00234 54* IF (DH.GT.0.0) GO TO 2
00236 55* 1 CONTINUE
00240 2 J = I-1
00241 57* DH = GH(J)-GEOHM/CF1
00242 58* TMK = T(J)-GT(J)*DH
00243 59* TATM = TMK*1.8
00244 10 CPATM = CPAIR(TATM)
00245 61* IF (ELEV .GT. 301000.) GO TO 15
00247 62* GAMATM = CPATM/(CPATM-0.0686)
00250 63* CATM = SORT(1.4*RU*TMK/WMO)/0.3048
00251 64* RETURN
00252 15 GAMATM = CPATM/(CPATM-(1.98585/MWT))
00253 66* CATM = SORT(GAMATM*RU*TMK/MWT)/0.3048
00254 67* RETURN
00255 20 Z = ELEV/(3280.83399)
00256 68* MWT = 28.9644-0.0309491*(Z-90.0)
00257 70* PZ = POLY(12,A,Z)
00260 71* PATM = PSL/(PZ**4)
00261 72* RZ = POLY(12,B,Z)
00262 73* RHOATM = RSL/(RZ**4)
00263 74* TATM = PATM*MWT/(RHOATM*1545.31)
00264 75* TMK = TATM/1.8
00265 76* GO TO 10
00266 77* END

```

END OF COMPILATION: NO DIAGNOSTICS.

\*\*\*\*\* ATMOS \*\*\*\*\*  
RHDS:P \*\*\*\*\* AVGMT/SCZ93 \*\*\*\*\*

DATE 071372

PAGE 3



\*\*\*\*\* AVGEMT/SCZ93 \*\*\*\*\*  
FOR S ME\*NASAS.AVGEMT/SCZ93,ME\*NASAS.AVGEMT/SCZ93  
FOR S9A-07/13/72-20:53:05 (0,)

SUBROUTINE AVGEMT ENTRY POINT 000110

STORAGE USED: CODE(1) 000121; DATA(0) 000040; BLANK COMMON(2) 000000

COMMON BLOCKS:

0000 AVGABS 000251

EXTERNAL REFERENCES (BLOCK, NAME)

0004 POLY  
0005 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000004 124G 0001 000024 132G 0000 R 000000 A 0000 I 000016 I  
0000 000025 INJPS 0000 I 000014 J 0004 R 000000 POLY 0000 R 000013 TALB 0003 R 000100 TFIN  
0000 R 000017 TFXP 0000 R 000015 TMPX 0003 R 000244 TPR 0000 R 000012 TSOL 0003 R 000062 XXFN  
0003 R 000000 XXVP 0003 R 000071 XXFN 0003 R 000031 XXXP

00101 1\* SUBROUTINE AVGEMT(I0)  
00101 2\* C  
00101 3\* C THIS SUBROUTINE COMPUTES:  
00101 4\* C THE NECESSARY VARIABLES FOR THE COMPUTATIONS OF THE AVERAGE  
00101 5\* C SURFACE COATING EVITTANCE  
00101 6\* C  
00103 7\* COMMON/AVGABS/XXMP(5,5),XXMP(5,5),XXFN(7),XXFN(7),  
00103 8\* TFIN(10,10),TPR(5)  
00104 9\* DIMENSION A(5), B(5)  
00105 10\* DATA A(1),A(2),A(3),A(4),A(5)/0.7804112E+00, -0.5527205E-04,  
00105 11\* 0.2530228E-06, -0.3229181E-09, 0.8854202E-13/  
00105 12\* B(1),B(2),B(3),B(4),B(5)/0.6538383E+00, 0.1144374E-03,  
00105 13\* -0.2432286E-07, -0.1437500E-09, 0.4947915E-13/  
00120 14\* DATA TSOL,TALB/10400.0,480.0/  
00123 15\* DO 1 J = 1,5  
00126 16\* TFPX = TPR(J)\*T0  
00127 17\* XXFN(J) = POLY(5/A,TFPX)  
00130 18\* XXXFN(J) = POLY(5/B,TFPX)  
00131 19\* DO 1 I = 1,5  
00134 20\* TFX = TFIN(J,I)\*T0  
00135 21\* XXMP(J,I) = POLY(5/A,TFNX)  
00136 22\* XXXMP(J,I) = POLY(5/B,TFNX)  
00141 23\* XXFN(6) = POLY(5/A,TALB)  
00142 24\* XXXFN(6) = POLY(5/B,TALB)  
00143 25\* XXFN(7) = 0.07156  
00144 26\* XXXFN(7) = 0.030562  
00145 27\* RETURN

\*\*\*\*\* AVGEMT/SCZ93 \*\*\*\*\*

00146 28\* END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG/P \*\*\*\*\* BETA/CFFC43 \*\*\*\*\*

\*\*\*\*\* BETA/CFFC43 \*\*\*\*\*  
QFOR,S ME\*NASAS.BETA/CFFC43,ME\*NASAS.BETA/CFFC43  
FOR S9A-07/13/72-20:53:07 (0.)

FUNCTION BETA ENTRY POINT 000015

STORAGE USED: CODE(1) 000017; DATA(0) 000010; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000000 BETA 0000 000003 INJPS 0000 R 000001 X1 0000 R 000002 X2

00101 1\* FUNCTION BETA(RHO,T)

00101 2\* C

00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :

00101 4\* C COEFFICIENT OF THERMAL EXPANSION AS A FUNCTION OF DENSITY

00101 5\* C (SLUG/QU.FT) AND TEMPERATURE (R) OF FC-43

00101 6\* C UNITS 1/R

00101 7\* C

00103 8\* DATA X1\*X2 /157.0883,-0.076167/

00106 9\* BETA = -X2/(X1+X2\*T)

00107 10\* RETURN

00110 11\* END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* BETA/CFFC75 \*\*\*\*\*

\*\*\*\*\* BETA/CF75 \*\*\*\*\*  
QFOR,S ME\*NASAS.BETA/CF75,ME\*NASAS.BETA/CF75  
FOR 59A-07/13/72-20:53:09 (0,)

FUNCTION BETA ENTRY POINT 000015

STORAGE USED: CODE(1) 000017; DATA(0) 000010; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000000 BETA 0000 000003 INJPS 0000 R 000001 X1 0000 R 000002 X2

00101 1\* FUNCTION BETA(RHO,T)

00101 2\* C

00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :

00101 4\* C COEFFICIENT OF THERMAL EXPANSION AS A FUNCTION OF DENSITY

00101 5\* C (SLUG/QU.FT) AND TEMPERATURE (R) OF FC-75

00101 6\* C UNITS 1/R

00101 7\* C

00103 8\* DATA X1\*X2 /155.522\*-0.085/

00106 9\* BETA = -X2/(X1+X2\*T)

00107 10\* RETURN

00110 11\* END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* BETA/CFHE \*\*\*\*\*

FOR S ME\*NASAS.BETA/CFHE,ME\*NASAS.BETA/CFHE  
FOR S9A-07/13/72-20:53:10 (0\*)

FUNCTION BETA ENTRY POINT 000106

STORAGE USED: CODE(1) 000112; DATA(0) 000037; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000005 A	0000 R 000014 ALPHA	0000 R 000002 A1	0000 R 000000 BETA	0000 R 000003 B1
0000 R 000004 C	0000 R 000017 DPDRHO	0000 R 000015 DPDT	0000 R 000016 DPDV	0000 000026 INJPS
0000 R 000001 R	0000 R 000006 RHOX	0000 R 000013 TM2	0000 R 000007 TX	0000 R 000010 V
0000 R 000011 VM2	0000 R 000012 VM3			

FUNCTION BETA(RHO,T)

00101 1\*  
 00101 2\*  
 00101 3\*  
 00101 4\*  
 00101 5\*  
 00101 6\*  
 00101 7\*  
 00101 8\*  
 00103 9\*  
 00111 10\*  
 00112 11\*  
 00113 12\*  
 00114 13\*  
 00115 14\*  
 00116 15\*  
 00117 16\*  
 00120 17\*  
 00121 18\*  
 00121 19\*  
 00122 20\*  
 00123 21\*  
 00124 22\*  
 00125 23\*

THIS FUNCTION SUBPROGRAM COMPUTES :  
 COEFFICIENT OF THERMAL EXPANSION AS A FUNCTION OF DENSITY  
 (SLUG/GU.FT) AND TEMPERATURE (R) OF HELIUM  
 UNITS 1/R

DATA R,A1,B1,C,A /2077.02,136.9595,3.5002295E-03,10.000658,1.49610  
 13E-02/  
 RHOX = RHO\*515.4275  
 TX = T/1.8  
 V = 1.0/RHOX  
 VM2 = RHOX\*RHOX  
 VM3 = RHOX\*VM2  
 TM2 = 1.0/(TX\*TX)  
 ALPHA = C\*RHOX/(TX\*\*3)  
 DPDT = R\*(V\*B1)\*VM2\*(1.0+2.0\*ALPHA)  
 DPDV = -R\*TX\*VM2\*(1.0+2.0\*RHOX\*B1)+A1\*VM3\*(2.0-3.0\*A\*RHOX)+R\*C\*VM3  
 1\*TM2\*(2.0+3.0\*B1\*RHOX)  
 DPDRHO = -DPDV/VM2  
 BETA = DPDT/(DPDRHO\*RHOX\*1.8)  
 RETURN  
 END

END OF COMPILATION: NO DIAGNOSTICS.

ENDG,P \*\*\*\*\* BETA/CFNAK \*\*\*\*\*

FOR S ME\*NASAS.BETA/CFNAK\*ME\*NASAS.BETA/CFNAK  
 FOR S9A-07/13/72-20:53:13 (0.)

FUNCTION BETA ENTRY POINT 000015

STORAGE USED: CODE(1) 000017; DATA(0) 000010; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000000 BETA 0000 000003 INJPS 0000 R 000001 X1 0000 R 000002 X2

00101	1*		FUNCTION BETA(RHO,T)
00101	2*	C	
00101	3*	C	THIS FUNCTION SUBPROGRAM COMPUTES :
00101	4*	C	COEFFICIENT OF THERMAL EXPANSION AS A FUNCTION OF DENSITY
00101	5*	C	(SLUG/GU.FT) AND TEMPERATURE (R) OF NAK 78.6
00101	6*	C	UNITS 1/R
00101	7*	C	
00103	8*		DATA X1,X2 /58.773064,-0.008433/
00106	9*		BETA = -X2/(X1*X2*T)
00107	10*		RETURN
00110	11*		END

END OF COMPILATION: NO DIAGNOSTICS.

END OF P \*\*\*\*\* BETA/CFSIL \*\*\*\*\*

FUNCTION BETA ENTRY POINT 000162

STORAGE USED: CODE(1) 000177; DATA(0) 000076; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 PF  
0004 NWDJ\$  
0005 NI02\$  
0006 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000	000030	IF	0000 R 000001 A1	0000 R 000022 A11	0000 R 000002 A2	0000 R 000003 A3
0000	R 000004	A4	0000 R 000005 A5	0000 R 000000 BETA	0000 R 000006 B1	0000 R 000023 B11
0000	R 000007	B2	0000 R 000010 B3	0000 R 000011 B4	0000 R 000012 B5	0000 R 000024 C
0000	R 000013	C1	0000 R 000025 C11	0000 R 000014 C2	0000 R 000017 DT	0000 000067 INJPS
0000	R 000026	P	0003 R 000000 PF	0000 R 000027 P1	0000 R 000020 THETA	0000 R 000021 THETA1
0000	R 000015	T0	0000 R 000016 T01			

FUNCTION BETA(RHO,T)

```

1* C
2* C
3* C THIS FUNCTION SUBPROGRAM COMPUTES :
4* C COEFFICIENT OF THERMAL EXPANSION AS A FUNCTION OF DENSITY
5* C (SLUG/CU.FT) AND TEMPERATURE (R) OF DOW CORNING 200
6* C SILICON OIL (1 CS)
7* C UNITS 1/R
8* C TEMPERATURE .6E. 359.67 AND .LE. 859.67
9* C
10* DATA A1,A2,A3,A4,A5 /12.35,2.98333,1.1,-0.48333,0.1/,B1,B2,B3,B4,B
11* /-1.5,-0.01333,-1.18,0.57333,-0.1/,C1,C2 /0.7767,-0.0288/,T0,T01
12* 2,DT /559.67,609.67,50.0/
13* THETA = (T-T0)/DT
14* THETA1 = (T-T01)/DT
15* A11 = ((4.0*A5*THETA*3.0*A4)*THETA+2.0*A3)*THETA+A2)*(1.0E-06)/DT
16* B11 = ((4.0*B5*THETA*3.0*B4)*THETA+2.0*B3)*THETA+B2)*(1.0E-09)/DT
17* C = C1+C2*THETA1
18* C11 = C2/DT
19* P = PF(RHO,T)
20* P1 = P/144.0-14.696
21* BETA= -C11/C-A11*P1-B11*P1*P1/2.0
22* IF (T.LT.360.67-OR.T.GT.860.67-OR.P.GT.146116.224-OR.P.GT.110116.2
23* 124.AND.T.LT.460.67) WRITE(6,1) T,P
24* 1 FORMAT (1H0,67HCOEFFICIENT OF THERMAL EXPANSION OF SILICON OIL, 0U
25* 1T OF RANGE, T = ,F10.5,6H, P = ,F15.5,/)
26* RETURN
27* END

```

END OF COMPILATION: NO DIAGNOSTICS.

DHDS,P \*\*\*\*\* CAPPA/CFFC43 \*\*\*\*\*



\*\*\*\*\* CAPPA/CFFC43 \*\*\*\*\*  
@FOR'S ME\*NASAS.CAPPA/CFFC43,ME\*NASAS.CAPPA/CFFC43  
FOR S9A-07/13/72-20:53:17 (0,)

FUNCTION CAPPA ENTRY POINT 000010

STORAGE USED: CODE(1) 000012; DATA(0) 000006; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000000 CAPPA 0000 000001 INJPS

00101	1*		FUNCTION CAPPA(RHO,T)
00101	2*	C	
00101	3*	C	THIS FUNCTION SUBPROGRAM COMPUTES :
00101	4*	C	ISOTHERMAL COMPRESSIBILITY AS A FUNCTION OF DENSITY
00101	5*	C	(SLUG/CU.FT) AND TEMPERATURE (R) OF FC-43
00101	6*	C	UNITS SQ.FT/LBF
00101	7*	C	
00103	8*		CAPPA = 0.0
00104	9*		RETURN
00105	10*		END

END OF COMPILATION: NO DIAGNOSTICS.

@HD6/P \*\*\*\*\* CAPPA/CFFC75 \*\*\*\*\*

\*\*\*\*\* CAPPA/CF75 \*\*\*\*\*

DFOR,S ME\*NASAS.CAPPA/CF75,ME\*NASAS.CAPPA/CF75  
FOR S9A-07/13/72-20:53:19 (0,)

FUNCTION CAPPA ENTRY POINT 00010

STORAGE USED: CODE(1) 000012; DATA(0) 000006; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000000 CAPPA 0000 000001 INJPS

00101	1*	C	FUNCTION CAPPA(RHO,T)
00101	2*	C	
00101	3*	C	THIS FUNCTION SUBPROGRAM COMPUTES :
00101	4*	C	ISOTHERMAL COMPRESSIBILITY AS A FUNCTION OF DENSITY
00101	5*	C	(SLUG/CU.FT) AND TEMPERATURE (R) OF FC-75
00101	6*	C	UNITS SQ.FT/LBF
00101	7*	C	
00103	8*	C	CAPPA = 0.0
00104	9*	C	RETURN
00105	10*	C	END

END OF COMPILATION: NO DIAGNOSTICS.

9H0G:P \*\*\*\*\* CAPPA/CFHE \*\*\*\*\*

\*\*\*\*\* CAPPA/CFHE \*\*\*\*\*

QFOR,S ME\*NASAS.CAPPA/CFHE\*ME\*NASAS.CAPPA/CFHE  
FOR 59A-07/13/72-20:53:21 (0,)

FUNCTION CAPPA ENTRY POINT 000064

STORAGE USED: CODE(1) 000067; DATA(0) 000031; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000005 A 0000 R 000002 A1 0000 R 000003 B1 0000 R 000004 C 0000 R 000000 CAPPA  
0000 R 000013 DPDV 0000 000022 INJPS 0000 R 000001 R 0000 R 000006 RHGX 0000 R 000012 TM2  
0000 R 000007 TX 0000 R 000010 VM2 0000 R 000011 VM3

00101 1\* FUNCTION CAPPA(RHO,T)

00101 2\* C

00101 3\* C

00101 4\* C

00101 5\* C

00101 6\* C

00101 7\* C

00103 8\* C

00103 9\*

00111 10\*

00112 11\*

00113 12\*

00114 13\*

00115 14\*

00116 15\*

00116 16\*

00117 17\*

00120 18\*

00121 19\*

THIS FUNCTION SUBPROGRAM COMPUTES :  
ISOTHERMAL COMPRESSIBILITY AS A FUNCTION OF DENSITY  
(SLUG/CU.FT) AND TEMPERATURE (R) OF HELIUM  
UNITS SQ.FT/LBF

DATA R,A1,B1,C,A /2077.02,136.9595,3.5002295E-03,10.000656,1.49610

13E-02/

RHOX = RHO\*515.4275

TX = T/1.8

VM2 = RHOX\*RHOX

VM3 = RHOX\*VM2

TM2 = 1.0/(TX\*TX)

DPDV = -R\*TX\*VM2\*(1.0+2.0\*RHOX\*B1)+A1\*VM3\*(2.0-3.0\*A\*RHOX)+R\*C\*VM3

1\*TM2\*(2.0+3.0\*B1\*RHOX)

CAPPA = -47.872\*RHOX/DPDV

RETURN

END

END OF COMPILATION: NO DIAGNOSTICS.

0M067P \*\*\*\*\* CAPPA/CFNAK \*\*\*\*\*

\*\*\*\*\* CAPPA/CFNAK \*\*\*\*\*  
 GFOR:5 ME\*NASAS.CAPPA/CFNAK,ME\*NASAS.CAPPA/CFNAK  
 FOR S9A-07/13/72-20:53:25 (0,)

FUNCTION CAPPA ENTRY POINT 000052

STORAGE USED: CODE(1) 000055; DATA(0) 000032; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3S

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000017 BETA	0000 R 000000 CAPPA	0000 R 000015 CP	0000 R 000010 C1	0000 R 000011 C2
0000 R 000007 DT	0000 000023 INJPS	0000 R 000016 RHOF	0000 R 000014 THETA	0000 R 000006 T0
0000 R 000020 VELS	0000 R 000012 V1	0000 R 000013 V2	0000 R 000001 X1	0000 R 000002 X2
0000 R 000003 X3	0000 R 000004 X4	0000 R 000005 X5		

00101 1\* FUNCTION CAPPA(RHO,T)

00101 2\* C  
 00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
 00101 4\* C ISOTHERMAL COMPRESSIBILITY AS A FUNCTION OF DENSITY  
 00101 5\* C (SLUG/CU.FT) AND TEMPERATURE (R) OF NAK 78.6  
 00101 6\* C UNITS SO.FT/LBF  
 00101 7\* C

00103 8\* DATA X1\*X2,X3,X4,X5 /0.2255,-0.016292,0.005396,-0.000758,0.000054/  
 00103 9\* 1,T0,DT /659.67,300.0/C1,C2 /58.773064,-0.008433/V1,V2 /6296.9267  
 00103 10\* 2,0.99/  
 00117 11\* THETA = (T-T0)/DT  
 00120 12\* CP = (((X5\*THETA+X4)\*THETA+X3)\*THETA+X2)\*THETA+X1  
 00121 13\* RHOF = C1+C2\*T  
 00122 14\* BETA = -C2/RHOF  
 00123 15\* VELS = V1+V2\*T  
 00124 16\* CAPPA = (32.174/(VELS\*VELS)+BETA\*BETA\*T/(CP\*778.26))/RHOF  
 00125 17\* RETURN  
 00126 18\* END

END OF COMPILATION: NO DIAGNOSTICS.

ENDG:P \*\*\*\*\* CAPPA/CFSIL \*\*\*\*\*

FOR S ME\*NASAS.CAPPA/CFSIL,ME\*NASAS.CAPPA/CFSIL  
FOR S9A-07/13/72-20:54:12 (07)

FUNCTION CAPPA ENTRY POINT 000132

STORAGE USED: CODE(1) 000146; DATA(0) 000064; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 PF  
0004 NWDUS  
0005 NIO2S  
0006 NERR3S

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000023 1F 0000 R 000016 A 0000 R 000001 A1 0000 R 000002 A2 0000 R 000003 A3  
0000 R 000004 A4 0000 R 000005 A5 0000 R 000017 B 0000 R 000006 B1 0000 R 000007 B2  
0000 R 000010 B3 0000 R 000011 B4 0000 R 000012 B5 0000 R 000000 CAPPA 0000 R 000014 DT  
0000 000055 INJPS 0000 R 000020 P 0003 R 000000 PF 0000 R 000021 P1 0000 R 000015 THETA  
0000 R 000022 TK 0000 R 000013 T0

00101 1\* FUNCTION CAPPA(RHO,T)

00101 2\*  
00101 3\* C  
00101 4\* C  
00101 5\* C  
00101 6\* C  
00101 7\* C  
00101 8\* C  
00101 9\* C  
00103 10\*  
00103 11\*  
00103 12\*  
00121 13\*  
00122 14\*  
00123 15\*  
00124 16\*  
00125 17\*  
00126 18\*  
00127 19\*  
00127 20\*  
00134 21\*  
00134 22\*  
00135 23\*  
00136 24\*

THIS FUNCTION SUBPROGRAM COMPUTES:  
ISOTHERMAL COMPRESSIBILITY AS A FUNCTION OF DENSITY  
(SLUG/CU.FT) AND TEMPERATURE (R) OF DOW CORNING  
200 SILICON OIL (1 CS)  
UNITS SG.FT/LBF  
TEMPERATURE .GE. 359.67 AND .LE. 859.67  
DATA A1,A2,A3,A4,A5 /12.35,2.90333,1.1,-0.48333,0.1/,B1,B2,B3,B4,  
185 /-1.5,-0.01333,-1.18,0.57333,-0.1/,T0,DT /559.67,50.0/  
THETA = (T-T0)/DT  
A = (((A5\*THETA+A4)\*THETA+A3)\*THETA+A2)\*THETA+A1)\*1.0E-06  
B = (((B5\*THETA+B4)\*THETA+B3)\*THETA+B2)\*THETA+B1)\*1.0E-09  
P = PF\*(RHO,T)  
P1 = P/144.0-14.696  
TK = A+B\*P1  
CAPPA = TK/144.0  
IF (T.LT.360.67.OR.T.GT.860.67.OR.P.GT.146116.224.OR.P.GT.110116.2  
124.AND.T.LT.460.67) WRITE(6,1) T,P  
1 FORMAT (1H0,6HISOTHERMAL COMPRESSIBILITY OF SILICON OIL, OUT OF R  
ANGE, T = ,F10.5,6H, P = ,F15.5//)  
RETURN  
END

END OF COMPILATION: NO DIAGNOSTICS.

\*\*\*\*\* CAPPA/CFSIL \*\*\*\*\*  
DNDG:P \*\*\*\*\* CNTLM \*\*\*\*\*

DATE 071372

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FOR S ME\*NASAS.CNTLM,ME\*NASAS.CNTLM  
 FOR S9A-0713/72-20:54:17 (07)

SUBROUTINE CNTLM ENTRY POINT 001771

STORAGE USED: CODE(1) 002060; DATA(0) 003423; BLANK COMMON(2) 001115

COMMON BLOCKS:

0003 SRTCNV 000003  
 0004 GRD 003670  
 0005 GIN 001610  
 0006 GEOM 000020  
 0007 FLDINL 000457  
 0010 DVCMFL 000002

EXTERNAL REFERENCES (BLOCK, NAME)

0011 THCFN  
 0012 CPFN  
 0013 DEFINI  
 0014 YINT  
 0015 HFL  
 0016 CPTB  
 0017 CPMP  
 0020 FINT  
 0021 SORT  
 0022 NERR2\$  
 0023 ALOG  
 0024 NWDU\$  
 0025 NI02\$  
 0026 NI01\$  
 0027 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000206	10L	0001	000471	100L	0001	000506	105L	0001	000162	147G	0001	000227	15L					
0001	000253	177G	0001	000110	2L	0001	000244	20L	0001	000347	230G	0001	000311	25L					
0001	000127	3L	0001	000332	30L	0001	000617	321G	0001	000632	330G	0001	000670	343G					
0001	000703	346G	0001	001061	374G	0001	001255	422G	0001	001274	426G	0001	001333	440G					
0001	001403	436G	0001	001437	464G	0001	000166	5L	0001	000340	50L	0001	000517	500L					
0001	000522	505L	0001	000542	506L	0001	000555	507L	0000	002504	509F	0000	002554	510F					
0001	001537	511G	0001	001002	513L	0001	001054	514L	0000	002605	515F	0000	003031	516F					
0001	001553	522G	0000	003057	530F	0001	001562	530G	0000	003064	535F	0001	001664	543L					
0001	001670	544L	0001	000405	55L	0000	003310	550F	0000	003317	555F	0000	003322	560F					
0001	001653	560G	0001	001176	600L	0001	000416	65L	0001	000147	7L	0001	000421	70L					
0000	R	002430	ASTR1	0000	R	002431	ASTR2	0000	R	000310	AUX1	0000	R	000322	AUX3				
0000	R	000334	AUX4	0000	R	000346	AUX5	0002	000752	COB	0002	R	002446	CONDMF					
0000	R	002454	COND1	0000	R	002455	COND2	0002	R	000574	CONFN	0000	R	000740	CONMP				
0012	R	000000	CPFN	0000	R	002467	CPFN1	0017	R	000000	CPMP	0000	R	000000	CPTB				
0000	R	002472	CPTB1	0002	001070	CPO	0004	003465	CSF	0000	R	002502	DEDOT	0000	R	002501	DEDT		
0013	R	000000	DEFINT	0002	001072	DELTA	0000	R	002464	DENTH	0000	R	002503	DIFE	0002	000764	D08		
0000	R	002354	DSTFN	0000	R	002412	DSTMP	0000	R	002400	DSTTB	0000	R	000360	DTDZ1	0000	R	000372	DTDZ2

\*\*\*\*\* CNTLM \*\*\*\*\*

```

0002 R 001110 DXI
0000 R 000000 DYI
0003 R 000000 ELEV
0004 R 003466 EXTSFN
0006 000016 FLYASS
0002 R 001040 FNU
0000 R 002451 FPRM1
0002 R 001036 FRE
0002 R 000024 FW
0002 R 001105 HFN
0002 I 002442 J
0002 I 001032 LIM
0002 I 001014 LL4
0000 I 002500 LSKIP
0004 003462 LTT
0002 I 000777 MZ
0002 I 001023 NEGJS
0002 001022 NRVP2
0005 001604 NTV
0006 R 000015 PLVASS
0005 001274 QITB
0000 R 002424 QSCLR
0002 R 001114 RHOFN
0006 R 000000 SFN
0000 R 002477 STORG
0006 R 000014 TBVASS
0000 R 002427 TPN2
0005 000000 TM
0004 000000 TRMTX
0005 001607 TX
0010 R 000000 WRAT
0002 000005 XIMP
0000 R 001370 XTI4E
0002 001055 XX3
0000 R 002433 ZE
0002 001052 DXITB
0004 003463 DZMFN
0000 R 002463 ENTHI
0002 R 001103 FENTH
0010 000001 FLUXI
0002 R 000012 FP
0002 R 001037 FR
0002 001043 FRL
0002 001112 FXOH
0002 001003 IFLOW
0002 001024 LC
0002 001011 LL1
0002 001016 LL6
0002 I 001005 LTB
0004 003460 MCVRD
0002 001034 NCCZ
0007 000000 NFLDTA
0002 001000 NRTB
0002 I 000776 NX
0002 R 001065 P0
0002 R 001102 GREF
0005 000764 GSTB
0002 R 001067 RH00
0004 003473 SS
0002 R 000017 STR
0002 R 000132 TEMP
0011 R 000000 THCFN
0002 R 002466 TT
0002 R 001064 T0
0000 R 002460 WWW
0002 000000 XITB
0002 001057 XX10
0002 001056 XX4
0002 R 000950 ZETA
0002 001052 DXITB
0004 003463 DZMFN
0000 R 002463 ENTHI
0002 R 001103 FENTH
0010 000001 FLUXI
0002 R 000012 FP
0002 R 001037 FR
0002 001043 FRL
0002 001112 FXOH
0002 001003 IFLOW
0002 001024 LC
0002 001011 LL1
0002 001016 LL6
0002 I 001005 LTB
0004 003460 MCVRD
0002 001034 NCCZ
0007 000000 NFLDTA
0002 001000 NRTB
0002 I 000776 NX
0002 R 001065 PHIF
0000 R 002445 GAERO
0002 R 000360 GRFN
0005 R 001606 GTO
0002 R 001073 RLIMIT
0000 R 002471 SFN
0000 R 002473 STTB
0000 R 002452 TFXN1
0002 001077 TI
0002 R 000276 TMP
0000 R 002457 TTT
0003 R 000001 VELS
0002 R 002447 WWX
0002 R 001106 XL
0002 001060 YINT
0014 R 000000 YINT
0000 R 002437 ZT
0004 003464 DX2
0004 003410 ERFN
0004 R 003467 EXTIFN
0007 R 000456 FLDINT
0000 R 002470 FNM
0000 R 000404 FPRME
0002 001111 FRD
0002 R 000036 FT
0015 R 000000 HFL
0000 003361 INJPS
0007 I 000001 LFLD
0002 001013 LL3
0002 001006 LMP
0002 001010 LTB2MZ
0002 I 001025 MSTOTR
0002 I 001026 NCTM
0002 001020 NRMP1
0006 I 000012 NTBS
0002 001071 PI
0000 R 002425 QIRED
0005 000144 QSFN
0002 R 001074 RDTWRT
0002 R 001100 RTEND
0000 R 002475 STMP
0003 R 000002 TATM
0000 R 002426 TFIN2
0002 001076 TINTL
0002 R 001075 TREF
0000 R 002440 TWRT
0002 000312 WIFLD
0007 R 000536 XIFN
0002 001054 XRE
0004 003446 XX2
0000 R 002436 ZCK
0004 003465 IM2
0002 001047 FZ
0000 I 002465 IM2
0002 001012 LL2
0002 001017 LL7
0002 001007 LTB2MZ
0002 I 001033 MOD
0002 001002 NCTL
0002 I 001001 NRPD
0005 001605 NSRD
0002 001062 PHIM
0005 000454 QIFN
0002 R 000524 GRMP
0000 R 002444 QTOT
0000 R 002461 RRR
0000 R 002366 STFNX
0000 R 002441 TABS
0000 R 002453 TFXN2
0006 R 000146 TIFLD
0002 R 000013 TIXL
0002 R 000050 TW
0000 R 002435 W
0002 R 001066 W0
0000 R 002476 XLTW
0002 001061 XX12
0000 R 002434 Z
0004 003465 ZZ2

```

SUBROUTINE CNTLM(Y,DY,DX,X,NTRY,IFVD)

THIS SUBROUTINE :

- 1- CONTROLS MAIN INTEGRATION
- 2- CONTROLS OUTPUT OF MAIN INTEGRATION
- 3- PROVIDES SYSTEM PERFORMANCE CHARACTERISTICS
- IS PART OF RKS AND DERIVM SUBROUTINES

```

COMMON XITB(5),XIMP(5),FP(10),FW(10),FT(10),TW(10.5),TEMP(10,10),
1 TMP(10.5),GRFN(10,10),GRMP(10),XIFN(10),ZETA(10),AUX2(10),
2 CONFN(10,10),CONMP(10),COS(10),DOB(10)
COMMON NX,MZ,NRTB,NRMP,NCTL,IFLOW,LT,LIBLMP,LTBMZ,LTB2MZ,LL1,
1 LL2,LL3,LL4,LL5,LL6,LL7,NR,API,NRTBI,NRMP2,NEQUS,LC,MSTOTR,
2 NCTM,NM1,MN1,LI,VRT,LI,MOD,NCCZ,NCONV
COMMON FRE,FR,FNU,FMACH,FPR,FRL,FRM,FFO,FFZ,FRAD,DZ,DXITB,DXIMP
1 XRE,XX3,XX4,XX10,XX11,XX12,PHIM,PHIF,T0,PO,W0,RH00,CPO,PI
2 DELTA,RLXMT,RTWRT,TREF,TINTL,TI,RTEND,FLUX,GREF,FENTH,
3 FREJ,HFN,XL,COH,DXI,FRD,FXOH,FXHW,RHOFN

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```

00101 1*
00101 2*
00101 3*
00101 4*
00101 5*
00101 6*
00101 7*
00101 8*
00101 9*
00103 10*
00103 11*
00104 12*
00104 13*
00104 14*
00105 15*
00105 16*
00105 17*
00105 18*

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00106 19* COMMON /SRTCNV/ ELEV,VELS,TATM
00107 20* COMMON /GRD/ TRMX(30,60),EBFN(5,5),EBMP(5),XX2(5),ZZ2(5),
00107 21* MCVRO,LCT,LT,DZMFN,
00107 22* DXX2,CSF,EXTSFN,EXTIFN,EXTSNP,EXTIMP,DXX21,SS(5,5,5)
00110 23* COMMON /GAIN/TM(100),OSFN(100,2),OJFN(100,2),OSTB(100,2),
00110 24* GIBB(100,2),NTM,NSRD,OTO, TX
00111 25* COMMON /GEOV/ SFN(10),NTBS,TNXL,TBMASS,PLMASS,FLMASS,STR
00112 26* COMMON /FLDINL/ NFDLTA,LFLD,TMEFLD(100),TIFLD(100),WIFLD(100),
00112 27* FLDINT
00113 28* COMMON /DVCVFL/ WRAT,FLUXI
00113 29*
00114 30* DIMENSION Y(200),DY(200),DY1(200),AUX1(10),AUX3(10),AUX4(10),
00114 31* AUX5(10),DIDZ1(10),DIDZ2(10),FPRME(500),XTIME(500)
00115 32* DIMENSION DSTFN(10),STFNX(10),DSTTB(10),DSTMP(10)
00115 33*
00116 34* QSOLR = EXTISFN*OTO
00117 35* QIRED = EXTIFN*OTO
00117 36*
00120 37* NCTM = NCTM+1
00120 38*
00121 39* TFN1 = TEMP(1,1)*TO
00122 40* TFN2 = TEMP(MZ,1)*TO
00123 41* ASTR1 = STR*SORT(THCFN(TFN1))*CPFN(TFN1)
00124 42* ASTR2 = STR*SORT(THCFN(TFN2))*CPFN(TFN2)
00125 43* IF (NCTM .GT. 500) GO TO 7
00125 44*
00127 45* IF (X .LT. XTIME(NCTM-1)) GO TO 2
00131 46* FPRME(NCTM) = ASTR1*DY(1)+ASTR2*DY(MZ)
00132 47* XTIME(NCTM) = X
00133 48* GO TO 3
00134 49* FPRME(NCTM) = FPRME(NCTM-1)
00135 50* XTIME(NCTM) = XTIME(NCTM-1)
00136 51* FPRME(NCTM-1) = ASTR1*DY(1)+ASTR2*DY(MZ)
00137 52* XTIME(NCTM-1) = X
00137 53*
00140 54*
00142 55* 3 IF (NCTM .GT. 1) GO TO 5
00143 56* FPRME(NCTM) = ASTR1*DY(1)+ASTR2*DY(MZ)
00143 57* XTIME(NCTM) = X
00144 58*
00146 59* 7 IF (NCTM .GT. 1) GO TO 5
00151 60* DO 4 I = 1,NEGUS
00153 61* 4 DY(I) = DY(I)
00154 62* GO TO 500
00156 63* 5 IF(IFVD.EQ.1) GO TO 10
00160 64* IF(NTY.EQ.3) GO TO 30
00160 65* GO TO (50,100),MSTOTR
00161 66*
00163 67* 10 IF(MSTOTR.EQ.1) GO TO 20
00164 68* ZE = X-RTEND
00166 69* IF(ABS(ZE)/ROTWRT.LE.0.0005) GO TO 65
00170 70* IF(ZE.GT.0.0) GO TO 105
00171 71* 15 IFVD = 1
00172 72* OX = ROTWRT
00174 73* IF(LIM.GT.LIMWRT) RETURN
00175 74* GO TO 500
00176 75* 20 Z = 0.0
DO 25 I = 1,NEGUS

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***** CNTLM *****
00201 76* IF(ABS(DY(I)).LT.1.0E-30) GO TO 25
00203 77* W = ABS(DY(I)/DY(I))
00204 78* IF(W.LE.0.0) GO TO 25
00206 79* ZCK = ALOG(W)
00207 80* IF(ABS(ZCK).LT.1.0E-06) GO TO 25
00211 81* Z = AMAX1(Z,ABS(DY(I)/ZCK)*DX)
00212 82* DY(I) = DY(I)
00214 83* IF(Z.LE.RLIMIT) GO TO 65
00216 84* IF(LIM.LE.LIMVRT) GO TO 15
00220 85* IFVD = 0
00221 86* RETURN
00222 87* 30 NTRY = 1
00223 88* IF(MOD.EG.1) GO TO 10
00225 89* GO TO 500
C
00225 90*
00225 91*
00226 92*
00227 93* DO 55 I = 1,NEOUS
00232 94* IF(ABS(DY(I)).LT.1.0E-30) GO TO 55
00234 95* W = ABS(DY(I)/DY(I))
00235 96* IF(W.LE.0.0) GO TO 55
00237 97* ZCK = ALOG(W)
00240 98* Z = AMAX1(Z,ABS(DY(I)/ZCK)*DX)
00242 99* DY(I) = DY(I)
00243 100* IF(Z.LE.RLIMIT) GO TO 65
00245 101* GO TO 70
00247 102*
00247 103*
00250 104*
00251 105* 65 NTRY = 2
00251 106* GO TO 505
C
00252 107*
00254 108* 70 IF(LIM.GT.LIMVRT) RETURN
00255 109* ZT = X*TWRT
00257 110* IF(ABS(ZT)/RDTWRT.LE.0.0005) GO TO 500
00261 111* NTRY = 3
00262 112* IF(DX.GE.RDTWRT) MOD = 1
00264 113* DX = DX*TWRT-X
00265 114* RETURN
C
00266 116*
00267 117* 100 ZE = X*RTEND
00271 118* IF(ABS(ZE)/RDTWRT.LT.0.0005) GO TO 65
00273 119* 105 NTRY = 3
00274 120* DX = DX*RTEND-X
00275 121* RETURN
C
00275 122*
00275 123*
00276 124*
00277 125* 500 TWRT = X*RDTWRT
00300 126* TABS = X*TREF
00302 127* IF(INCONV.GT.0) GO TO 506
00310 129* WRITE(6,510) TABS,X,NCTM
00310 130* GO TO 507
00310 131* 506 WRITE(6,509) TABS,X,NCTM,ELEV,VELS,TATM
C
00320 132*
00320 133* 507 DO 508 J=1,MZ

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00323 133* FT(J) = Y(LT+J)
00324 134* TW(J,1) = Y(LTB+J)
00325 135* TMP(J,NRMP) = Y(LL5+J)
00326 136* TEMP(J,1) = Y(LL4+J)
00327 137* DO 508 I=2,NX
00332 138* K = MZ*(I-2)+J
00333 139* 508 TEMP(J,I) = Y(K)
00336 140* IF (LFLD.EQ. 2) FT(1) = FLOINT
00336 141*
00336 142*
00340 143* 509 FORMAT(1H1,5X,20HELAPSED TIME IS ,F12.4,6H HR , ,
00340 144* 1 6X,20HRELATIVE TIME IS ,F12.4,
00340 145* 2 6X,I7,14H INTEGR. STEPS,/,
00340 146* 3 6X,20HALTITUDE IS ,F10.2,8H FT , ,
00340 147* 4 6X,20HVELOCITY IS ,F10.2,10H FT/SEC , ,
00340 148* 5 6X,22HATM. TEMPERATURE IS ,F8.2,2H R,/)
00341 149* 510 FORMAT(1H1,5X,25HELAPSED TIME IS ,F12.4,
00341 150* 1 6X,25HRELATIVE TIME IS ,F12.4,
00341 151* 2 18X,12H*IN ORBIT**20X,I7,14H INTEGR. STEPS,/)
00341 152*
00341 153*
00341 154*
00342 155* LISTING OF FIN TEMPERATURE DISTRIBUTION
00342 156*
00342 157* DO 512 J=1,MZ
00350 158* DO 511 I=1,NX
00351 159* AUX4(I) = CONFN(J,I)
00353 160* AUX5(J) = DEFINT(AUX4,DXI,NX)+CONVP(J)*COH
00354 161* 512 AUX3(J) = DEFINT(AUX1,OXI,NX)+GRWP(J)*COH
00356 162* GTOT = DEFINT(AUX3,DZ,MZ)*GREF*XL
00357 163* GAERO = DEFINT(AUX5,DZ,MZ)*GREF*XL
00357 164*
00357 165* CONDUCTION FROM MANIFOLD
00357 166*
00360 167* IF (NCTM .GT. 500) GO TO 514
00362 168* IF (NCTM .GT. 3) GO TO 513
00365 169* CONDMF = 0.66667*FPRME(1)*SQRT(XTIME(NCTM))
00366 170* GO TO 600
00367 171* 513 WWX = X/3.0
00370 172* FPRM2 = YINT(XTIME,FPRME,NCTM,3,WWX)
00371 173* WWX = 2.0*WWX
00372 174* FPRM1 = YINT(XTIME,FPRME,NCTM,3,WWX)
00372 175* CONDMF = SQRT(XTIME(NCTM))/105.0*(68.0*FPRME(NCTM)+90.0*FPRM1+
00372 176* 1 36.0*FPRM2+16.0*FPRME(1))
00373 177*
00376 178* 514 DO 580 I=1,NX
00377 179* TFNX1 = TEMP(1,I)*T0
00377 180* DTDZ1(I) = -THCFN(TFNX1)*SFN(I)*(-3.0*TEMP(1,I)+4.0*TEMP(2,I)
00400 181* -TEMP(3,I))
00401 182* 1 TFNX2 = TEMP(MZ,I)*T0
00401 183* 580 DTDZ2(I) = THCFN(TFNX2)*SFN(I)*(3.0*TEMP(MZ,I)-4.0*TEMP(MZ-1,I)+
00403 184* 1 COND1 = DEFINT(OTDZ1,OXI,NX)
00404 185* COND2 = DEFINT(OTDZ2,OXI,NX)
00405 186* CONMF1 = (COND1+COND2)*NTBS*TO*HFV/(DZ*XL)
00406 187* IF (CONDMF .LE. CONMF1 .OR. NCTM .GT. 500) CONDMF = CONMF1
00406 188*
00406 189* C FLUID ENTHALPY REJECTION

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\*\*\*\*\* CNTLM \*\*\*\*\*

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00406 190* C 600 TTT = FT(1)*T0
00410 191* = FW(1)*W0
00411 192* RRR = RHOD/FW(1)
00412 193* ENTHI = FLUX*(HFL(RRR,TTT)*WWW**2/1556.36)*WRAT
00413 194* TTT = FT(MZ)*T0
00414 195* WWW = FW(MZ)*W0
00415 196* RRR = RHOD/FW(MZ)
00416 197* ENTHE = FLUX*(HFL(RRR,TTT)*WWW**2/1556.36)*WRAT
00417 198* DENTH = ENTHI-ENTHE
00420 199* C ENERGY STORAGE RATE
00420 201* C FIN
00420 202* C DO 700 I=2,NX
00420 203* C IM2 = I-2
00420 204* C DO 700 J=1,MZ
00421 205* K = IM2*MZ+J
00422 206* TT = T0*TEMP(J,I)
00423 207* CPFN1 = CPFN(TT)
00424 208* DSTFN(J) = CPFN1*SFNI(I)*DY(K)
00425 209* STFNX(I) = DEFINT(DSTFN,DZ,MZ)
00430 210* C DO 705 J=1,MZ
00431 211* K = LTB+J
00432 212* TT = T0*TEMP(J,I)
00433 213* CPFN1 = CPFN(TT)
00434 214* DSTFN(J) = CPFN1*DY(K)
00435 215* STFNX(1) = DEFINT(DSTFN,DZ,MZ)*SFN(1)
00436 216* FNM = 2.0*RHO*FN*TNXL*HFN
00437 217* STFN = DEFINT(STFNX,DXI,NX)*FNM
00442 218* C TUBE
00443 219* C DO 710 J=1,MZ
00444 220* K = LTB+J
00445 221* TT = T0*TEMP(J,I)
00446 222* CPFN1 = CPFN(TT)
00447 223* DSTFN(J) = CPFN1*DY(K)
00448 224* STFNX(1) = DEFINT(DSTFN,DZ,MZ)*SFN(1)
00449 225* FNM = 2.0*RHO*FN*TNXL*HFN
00450 226* STFN = DEFINT(STFNX,DXI,NX)*FNM
00451 227* C PROTECTION LAYER
00452 228* C DO 715 J=1,MZ
00453 229* K = LTB+J
00454 230* TT = T0*TEMP(J,I)
00455 231* CPMP1 = CPMP(TT)
00456 232* DSTMP(J) = CPMP1*DY(K)
00457 233* STMP = DEFINT(DSTMP,DZ,MZ)*PLMASS
00458 234* C XLTW = XL/(T0*W0*3600.0)
00459 235* STORG = (STFN*STTB+STMP)/XLTW
00460 236* C STORG = DENTH+CONDMF+QAERO-QTOT
00461 237* C WRITE(6,515) T0,QSOLR,QREF,QIRED,QTOT,QAERO,CONDMF,STORG,
00462 238* C 1(XIFN(I),I=1,NX)
00463 239* C 515 FORMAT(1H0,15X,55HF1N TEMPERATURE DISTRIBUTION AND RADIANT HEAT RE
00464 240*
00465 241*
00466 242*
00467 243*
00468 244*
00469 245*
00470 246*
00471 247*
00472 248*
00473 249*
00474 250*
00475 251*
00476 252*
00477 253*
00478 254*
00479 255*
00515 256*

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00515 247* 1JECTION,,16X,55H*****
00515 248* 2*****,,
00515 249* 3 2X,33REFERENCE TEMPERATURE, T00 =,F10,3,13H R ,/,
00515 250* 4 2X,56HREF. RADIANT HEAT FLUX ,
00515 251* 5 8X,33INCIDENT SOLAR FLUX, QSOLR. =,E10,4,15H BTU/(HR SQ FT),
00515 252* 6 /,
00515 253* 7 2X,33H PER UNIT AXIAL LENGTH, QREF =,E10,4,13H BTU/(HR FT),
00515 254* 8 8X,33INCIDENT INFRARED FLUX, QIRED =,E10,4,15H BTU/(HR SQ FT),
00515 255* 9 /,
00515 256* 1 2X,33HTOT. RADIANT REJECTION QTOT =,E10,4,13H BTU/HR ,
00515 257* 2 8X,33HAERODYN. HEATING POWER, QCONV =,E10,4,7H BTU/HR,,/
00515 258* 3 2X,33HCOND. FROM MANIFOLDS, CONDMF =,E10,4,13H BTU/HR ,
00515 259* 4 8X,33HEMERY STORAGE RATE, STORG =,E10,4,7H BTU/HR,,/
00515 260* 5/,6H AXIAL,4X,8HRELATIVE,14X,30HRELATIVE TEMPERATURE OF FIN, T,,/6
00515 261* 6H DIST.,4X,9HRAD. HEAT,,/10X,9HREJECTION,15X,33HDISTANCE NORMAL TO
00515 262* 7 FLOW DIRECTION,,/3H Z,9X,1H0,31X,1HX,,/20X,10F9,5,1
00515 263* WRITE(6,516)
00520 264* 516 FORMAT(1H,120H)
00520 265* 1-----
00520 266* 2----)
00521 267* DO 525 J=1,MZ
00524 268* 525 WRITE(6,530) ZETA(J),AUX3(J),(TEMP(J,I),I=1,NX)
00535 269* 530 FORMAT(1H,1F5,3F11,4,2X,10F9,4)
00536 270* WRITE(6,555)
00536 271*
00540 272* C
00551 273* WRITE(6,535) T0,P0,W0,FENTH,ENTHI,ENTHE,ODENTH
00551 274* 10ROID PROTECTION LAYER,,/16X,70H*****
00551 275* 2*****
00551 276* 3 10X,33REFERENCE TEMPERATURE, T00 =,F15,3,12H R ,/,
00551 277* 4 10X,33REFERENCE PRESSURE, P00 =,F15,3,12H LRF/SQ.FT ,/,
00551 278* 5 10X,33REFERENCE VELOCITY, V00 =,F15,3,12H FT/SEC ,/,
00551 279* 6 6X,36HCOOLANT POWER, INLET AT T=10 H0 = ,E11,6,7H BTU/HR,,/
00551 280* 7 6X,36H INLET CURRENTLY HI = ,E11,6,7H BTU/HR,,/
00551 281* 8 6X,36H EXIT CURRENTLY EI = ,E11,6,7H BTU/HR,,/
00551 282* 9 6X,36H TOT. REJECTION DH = ,E11,6,7H BTU/HR,,/
00551 283* 9 58X,8HPROTECT.,10X,18HENTHALPY REJECTION,,/36X,5HFLUID,7X,4HWALL,
00551 284* 17X,5HLAYER,7X,13HPER UNIT TUBE,5X,8HFRACTION,,/6H AXIAL,5X,8HPRESS
00551 285* 2URE,4X,8HVELOCITY,9X,23HT E M P E R A T U R E S,12X,6HLENGTH,11X,2
00551 286* 3-HOF,,/6H DIST.,8X,1HP,11X,1HW,11X,2HTF,9X,3HTWI,9X,3HTMP,9X,
00551 287* 411HBTU/(HR FT),7X,5HTOTAL,,/
00551 288* C
00551 289* C
00551 290* POWER FLUX, FLUID FLOW
00552 291* LSKIP = 1
00553 292* CALL FINT(AUX2,0,0,DZ,WZ,AUX1)
00554 293* DEDT = AUX1(1)-AUX1(MZ)
00555 294* IF(ABS(DEDT).LT.1.0E-08) LSKIP = 2
00557 295* DO 545 J=1,MZ
00562 296* DEDOT = 0.0
00563 297* GO TO (543,544), LSKIP
00564 298* 543 DEDOT = (AUX1(1)-AUX1(J))/DEDT
00565 299* 544 DIFE = AUX2(J)*FREQ
00566 300* 545 WRITE(6,550) ZETA(J),FP(J),FW(J),FT(J),TW(J,1),TMP(J,NRMP),DIFE,
00601 301* DEDOT
00601 302* 550 FORMAT(1H,1F5,3F12,5F11,4,3F12,4,F17,6F14,5)
00602 303* WRITE(6,555)

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\*\*\*\*\* CNTLM \*\*\*\*\*

```
00604 304* 555 FORMAT(1H,7H (EXIT))
00604 305* C
00605 306* LIM = LIM+1
00606 307* IF (MSTOTR.EQ.1 .AND. NTRY.EQ.2) WRITE (6,560)
00611 308* 560 FORMAT (1H1,25H STEADY STATE IS REACHED)
00612 309* RETURN
00613 310* END
```

END OF COMPILATION: NO DIAGNOSTICS.

SHDG,P \*\*\*\*\* CNTLN \*\*\*\*\*

\*\*\*\*\* CNTLN \*\*\*\*\*

QFOR,S ME\*NASAS.CNTLN,ME\*NASAS.CNTLN  
FOR S9A-07/13/72-20:54:26 (0,)

SUBROUTINE CNTLN ENTRY POINT 000075

STORAGE USED: CODE(1) 000111; DATA(0) 000023; BLANK COMMON(2) 001115

COMMON BLOCKS:

0003 DRVLCM 000024  
0004 DRVLCN 000001

EXTERNAL REFERENCES (BLOCK, NAME)

0005 NWDJ\$  
0006 NI01\$  
0007 NI02\$  
0010 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000	000001	IF	0001	00025	1206	0002	000562	AUX2	0004	R	000000	B2	0002	000752	COB
0002	001107	COH	0002	000574	CONFN	0002	000740	CONMP	0002	001070	CP0	0002	0002	001072	DELTA
0002	000764	DOE	0002	001110	DXI	0002	001053	DXIMP	0002	001052	DXITB	0002	0002	001051	DZ
0002	001103	FEMTH	0002	001045	FFO	0002	001101	FLUX	0002	001041	FMACH	0002	0002	001040	FNU
0002	001046	FOF	0002	R	000012	FP	0002	001042	FPR	0002	001037	FR	0002	001050	FRAD
0002	001111	FRD	0002	001036	FRE	0002	001104	FREJ	0002	001043	FRL	0002	0002	001044	FRM
0002	R	000036	FT	0002	R	000024	FW	0002	001113	FXOH	0002	001112	0002	001047	FZ
0002	001105	HFN	0000	I	000000	I	0002	I	001003	IFLOW	0000	000007	0002	001024	LC
0002	001032	LIN	0002	001031	LIMVRT	0002	001011	LL1	0002	001012	LL2	0002	0002	001013	LL3
0002	001014	LL4	0002	001015	LL5	0002	001016	LL6	0002	001017	LL7	0002	0002	001006	LMP
0002	001004	LT	0002	001005	LTB	0002	001007	LTBMZ	0002	001010	LTBMZ	0002	0002	001030	VW1
0002	001033	MOC	0002	001025	MSOTR	0002	001027	MZ	0002	001034	NCCZ	0002	0002	001035	NCONV
0002	I	001002	NCTL	0002	001026	NCT4	0002	001023	NEQUS	0002	001021	NRTB1	0002	001001	NRMP
0002	001020	NRAP1	0002	001022	NRMP2	0002	001000	NRTB	0002	001021	NRTB1	0002	0002	000776	NX
0002	001063	PHIF	0002	001062	PHIM	0002	001071	PI	0002	001065	P0	0002	0002	001102	REF
0002	000360	GRFN	0002	000524	GRMP	0002	001074	ROTWT	0002	001114	RHOFN	0002	0002	001067	RHO0
0002	001073	RLIMIT	0002	001100	RTEND	0002	R	000000	STRGE1	0003	R	000012	0002	000132	TEMP
0002	001077	TI	0002	001076	TINTL	0002	000276	TMP	0002	001075	TREF	0002	0002	000050	TW
0002	001064	T0	0002	001066	W0	0002	000536	XIFN	0002	000005	XITP	0002	0002	000000	XITB
0002	001106	XL	0002	001054	XRE	0002	001057	XX10	0002	001060	XX11	0002	0002	001061	XX12
0002	001055	XX3	0002	001056	XX4	0002	000550	ZETA							

00101	1*	C	SUBROUTINE CNTLN (Y,DY,DX,X,NTRY,IFVD)
00101	2*	C	
00101	3*	C	THIS SUBROUTINE CONTROLS :
00101	4*	C	1- INTEGRATION OF COOLANT FLUID PRESSURE AND VELOCITY FIELDS
00101	5*	C	2- INTEGRATION FOR FLUID INITIAL CONDITIONS
00101	6*	C	3- PART OF RK5F AND DERIVL SUBROUTINES
00101	7*	C	

\*\*\*\*\* CNTLN \*\*\*\*\*

```

00103 8* COMMON XITB(5),XIMP(5),FP(10),FW(10),FT(10),TW(10,5),TEMP(10,10),
00103 9* TMP(10,5),GRFN(10,10),ORMP(10),XIFN(10),ZETA(10),AUX2(10),
00103 10* CONFN(10,10),CONMP(10),COB(10),DOB(10)
00104 11* COMMON NX,MZ,NRTB,NRMP,NCTL,IFLOW,LI,LTB,LMP,LTBMZ,LTB2MZ,LL1,
00104 12* LL2,LL3,LL4,LL5,LL6,LL7,NRMP1,NRTB1,NRMP2,NEGUS,LC,MSTOIR,
00104 13* NCTM,NMI,MVI,LI,MVRT,LIM,MOD,NCCZ,NCONV
00105 14* COMMON FRE,FR,FNU,FMACH,FFR,FRL,FRM,FFO,FOF,FZ,FRAD,DZ,DXITB,DXIMP
00105 15* ,XRE,XX3,XX4,XX10,XX11,XX12,PHIM,PHIF,T0,P0,W0,RH00,CP0,PI
00105 16* ,DELTA,RLIMIT,PROTWR,TREF,TINTL,TI,RTEND,FLUX,REF,FENTH,
00105 17* FREJ,HFN,XL,COH,DXI,FRD,FXOH,FXHW,RHOFN
00106 18* COMMON /DRVLCM/ STRGE1(10),STRGE2(10)
00107 19* COMMON /DRVLCM/ B2
00110 20* DIMENSION Y(3),DY(3)
00110 21* C
00111 22* STRGE1(NCTL) = B2
00112 23* STRGE2(NCTL) = DY(3)
00113 24* IF(IFLOW.EQ.1) WRITE (6,1) NCTL,X,(Y(I),I=1,3),TW(NCTL,1)
00125 25* IF(NCTL.EQ.MZ) NTRY=2
00127 26* FP(NCTL) = Y(1)
00130 27* FW(NCTL) = Y(2)
00131 28* IF (IFLOW.EQ. 1) FT(NCTL) = Y(3)
00133 29* NCTL = NCTL+1
00134 30* RETURN
00134 31* C
00135 32* 1 FORMAT (15X,15,F10.3,4F20.6)
00136 33* END

```

END OF COMPILATION: NO DIAGNOSTICS.

PH06,P \*\*\*\*\* CONVEC \*\*\*\*\*



\*\*\*\*\* CONVEC \*\*\*\*\*

QFOR,S ME\*NASAS5.CONVEC,ME\*NASAS5.CONVEC  
FOR 59A-07/13/72-20:54:33 (0,)

SUBROUTINE CONVEC ENTRY POINT 000543

STORAGE USED: CODE(1) 000561; DATA(0) 000144; BLANK COMMON(2) 001115

COMMON BLOCKS:

0003 VELALT 001132  
0004 CNV 000010  
0005 SRTCNV 000003  
0006 GRD 003721  
0007 SSF 000005

EXTERNAL REFERENCES (BLOCK, NAME)

0010 ALTVEL  
0011 ATMOS  
0012 ENTAIR  
0013 TNH  
0014 CPAIR  
0015 REFP  
0016 NUS  
0017 NEXP6\$  
0020 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000031	122G	0001	000033	1256	0001	000425	150L	0001	000401	211G	0001	000416	220G														
0001	000417	223G	0001	000431	231G	0001	000507	243G	0001	000510	246G	0001	000122	31L														
0001	000236	35L	0001	000253	37L	0001	000267	47L	0001	000045	5L	0001	000277	51L														
0002	001105	AL	0007	000002	ALPFN	0003	R	000310	ALTA	0003	R	000454	ALTR	0010	R	000000	ALTVEL											
0002	000562	AUX2	0000	R	000020	CATM	0002	000752	COB	0000	R	000043	COEFF	0004	000007	COEFF	0000	R	000054	CPFIN								
0002	001107	COH	0002	R	000574	CONFN	0002	R	000740	CONMP	0014	R	000000	CPAIR	0000	R	000764	DOB	0002	000764	DOB							
0004	000003	DX	0002	001110	DXI	0002	001053	DXIMP	0002	001072	DELTA	0002	001052	DXITB	0006	003464	DX2	0006	003464	DX2	0006	003464	DX2					
0006	003472	DX21	0002	001051	DZ	0006	003463	DZMFN	0006	003410	ERFN	0006	003441	EBMP	0006	003441	EBMP	0000	R	000030	ENTFN	0000	R	000030	ENTFN			
0005	R	000000	ELEV	0012	R	000000	ENTAIR	0000	R	000021	ENTATM	0006	000042	ENTAW	0000	R	000030	ENTFN	0000	R	000030	ENTFN	0000	R	000030	ENTFN		
0006	003467	EXTIFN	0006	003471	EXTIMP	0006	003466	EXTSFN	0006	003470	EXTSMP	0006	003470	EXTSMP	0002	001103	FENTH	0002	001103	FENTH	0002	001103	FENTH	0002	001103	FENTH		
0002	001045	FFO	0002	001101	FLUX	0002	001041	FMACH	0002	001037	FR	0002	001050	FRAD	0002	001111	FRD	0002	001111	FRD	0002	001111	FRD	0002	001111	FRD		
0002	000012	FP	0002	001104	FREJ	0002	001043	FRL	0002	001044	FRM	0002	001044	FRM	0002	000036	FT	0002	000036	FT	0002	000036	FT	0002	000036	FT		
0002	000024	FW	0002	001113	FXHW	0002	001112	FXOH	0002	001112	FXOH	0002	001047	FZ	0002	001105	HFN	0002	001105	HFN	0002	001105	HFN	0002	001105	HFN		
0000	I	000051	I	0002	001003	IFLOW	0000	I	000024	II	0000	I	000016	INJPS	0004	I	000002	IOPTN	0004	I	000002	IOPTN	0004	I	000002	IOPTN		
0000	I	000050	J	0000	I	000025	JJ	0000	I	000053	L	0002	001024	LC	0006	003461	LCT	0006	003461	LCT	0006	003461	LCT	0006	003461	LCT		
0002	001032	LIM	0002	001031	LIMWRT	0002	001011	LL1	0002	001011	LL1	0002	001012	LL2	0002	001013	LL3	0002	001013	LL3	0002	001013	LL3	0002	001013	LL3		
0002	001014	LL4	0002	001015	LL5	0002	001016	LL6	0002	001017	LL7	0002	001017	LL7	0002	001006	LMP	0002	001006	LMP	0002	001006	LMP	0002	001006	LMP		
0002	001004	LT	0002	001005	LTB	0002	001007	LTRMZ	0002	001010	LTRMZ	0002	001010	LTRMZ	0006	003462	LTT	0006	003462	LTT	0006	003462	LTT	0006	003462	LTT		
0000	I	000026	M	0000	R	000000	MACHNO	0006	I	003460	MCVRD	0000	I	000017	MM	0002	001030	MM1	0002	001030	MM1	0002	001030	MM1	0002	001030	MM1	
0002	001033	M0D	0002	001025	MSTOTR	0000	R	000002	MWT	0000	R	000002	MWT	0002	000777	MZ	0003	I	001130	NA	0003	I	001130	NA	0003	I	001130	NA
0002	001034	NCCZ	0002	001035	NCONV	0002	001002	NCTL	0002	001002	NCTL	0002	001026	NCTM	0002	001023	NEGUS	0002	001023	NEGUS	0002	001023	NEGUS	0002	001023	NEGUS		
0002	001027	NM1	0000	I	000016	NN	0003	I	001131	NR	0002	I	001001	NRMP	0002	I	001020	NRMPI	0002	I	001020	NRMPI	0002	I	001020	NRMPI		



\*\*\*\*\* CONVEC \*\*\*\*\*

```

00136 38* ENTATM = ENTAIR(TATM)
00137 39* MACHNO = VEL5/CATH
00140 40* REFRP = 0.72
00141 41* REFGAM = 1.3
00142 42* I1 = MZ/2
00143 43* JJ = NX/2
00144 44* M = 1
00145 45* TMPE = TEMP(II,JJ)*TIN
00146 46* ENTFN = ENTAIR(TMPE)
00147 47* RECFAC = REFRP*((1.0+0.528*MACHNO**2)/(22.0+MACHNO**2))
00150 48* REFENT = 0.5*(ENTATM+ENTFN)+0.11*RECFAC*
1 REFGAM-1.0)*MACHNO**2*ENTATM
00151 49* REFT(M) = TNH(REFENT)
00152 51* REFTP = REFT(M)
00153 52* TMK = REFTP/1.8
00154 53* REFRVIS = ((1.458E-06*TMK**1.5)/(TMK+110.4))*0.671969
00155 54* KEFK = ((6.325E-07*TMK**1.5)/(TMK+245.4*10.0**(-12.0/TMK)))
1 *0.671969
00156 56* REFCP = CPAIR(REFTP)
00157 57* REFRP = REFRVIS*REFCP/REFK
00160 58* IF (ELEV .GT. 301000.) GO TO 35
00162 59* REFGAM = REFCP/(REFCP-0.06806)
00163 60* GO TO 37
00164 61* MWT = ELEV/(3280.8399)
00165 62* REFGAM = REFCP/(REFCP-(1.98585/MWT))
00166 63*
00167 64* 37 IF (M.EG. 1) GO TO 47
00171 65* IF (ABS(REFT(M))-REFT(M-1)) .LE. 5.0) GO TO 51
00173 66* 47 M = M+1
00174 67* IF (M .GT. 10) GO TO 51
00176 68* GO TO 31
00177 69* 51 CONTINUE
00200 70* CALL REFP (ELEV,REFTP,REFPR,REFVIS,REFRHO,REFK,REFCP,
00200 71* REFGAM)
00201 72* ENTAW = ENTATM+RECFAC*VELS**2/50062.744
00202 73* CALL NUS(MACHNO,TATM,CATH,TIN,REFPR,REFVIS,REFRHO,
00202 74* STAGX,VERTX,NSRAD,NUS1)
00203 75* COEF = (3600.*REFK)/(STAGX*REFCP)
00203 76*
00204 77* C DETERMINATION OF CONVECTION LOSS FROM METEOROID PROTECTION NODES
00203 78* C
00204 79* TMET = TMP(II,NRMP)*TIN
00205 80* CPMP = CPAIR(TMET)
00206 81* XXX = COEF*NUS1
00207 82* YYY = CPMP*TIN
00210 83* DO 80 J = 1,MZ
00213 84* 80 CONMP(J) = XXX*(ENTAW-YYY*TMP(J,NRMP))/(ST4*2.)
00213 85* C
00213 86* C DETERMINATION OF FRACTION OF FIN NODES COVERED BY PROTECTION LAYER
00213 87* C
00215 88* IF (MCVRD .EQ. 0) GO TO 150
00222 90* DO 140 I = 1,MCVRD
00222 90* DO 140 J = 1,MZ
00225 91* 140 CONFN(J,I) = 0.0
00230 92* DO 160 J = 1,MZ
00233 93* IFIN = TEMP(J,MCVRD+1)*TIN
00234 94* ENTFN = ENTAIR(TFIN)

```

```

***** CONVEC *****
00235 95* 160 CONFN(J,MCVRD+1) = XSPM*COEF*NUS1*(ENTAW-ENTFN)/ST4
00237 96* L = MCVRD+2
00237 97*
00237 98* C DETERMINATION OF CONVECTION LOSS FROM FIN NODES
00237 99* C
00240 100* CFFIN = CPAIR(TMPE)
00241 101* ZZZ = CFFIN*TIM
00242 102* DO 180 I = L,NX
00245 103* DO 180 J = 1,MZ
00250 104* 180 CONFN(J,I) = XXX*(ENTAW-ZZZ*TEMP(J,I))/ST4
00253 105* 1000 RETURN
00254 106* END

```

END OF COMPILATION: NO DIAGNOSTICS.

GMDS:P \*\*\*\*\* CPAIR \*\*\*\*\*

CPAIR,ME\*NASAS.CPAIR,ME\*NASAS.CPAIR  
FOR S9A-07/13/72-20:54:38 (0,)

FUNCTION CPAIR ENTRY POINT 000074

STORAGE USED: CODE(1) 000110; DATA(0) 000035; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 POLY  
0004 SORT  
0005 NERRJ3

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000054 10L 0001 000016 5L 0000 R 000001 A 0000 R 000012 CPN  
0000 R 000013 CPO 0000 R 000007 FMN 0000 R 000006 FMO 0000 000027 INJPS 0003 R 000000 POLY  
0000 R 000011 WMN 0000 R 000010 WHO

00101 1\* FUNCTION CPAIR(T)

00101 2\* C  
00101 3\* C  
00101 4\* C  
00101 5\* C  
00101 6\* C  
00101 7\* C

THIS FUNCTION SUBPROGRAM COMPUTES:  
THE SPECIFIC HEAT OF AIR /S A FUNCTION OF TEMPERATURE (R)  
UNITS BTU/(LBM.R)

DIMENSION A(5)

DATA FMO,FMN,WMO,WMN / 0.234559 ,0.765441,31.9988,28.0134/  
DATA A(1),A(2),A(3),A(4),A(5)/0.34240E+00,-0.95225E-03,  
1 0.31862E-05,-0.45750E-08,0.23750E-11/  
IF (T .GT. 600.) GO TO 5

CPAIR = POLY (5,A,T)  
RETURN

5 CPN = 9.47-3470./T+1160000./T\*\*2  
CPO = 11.151-172./SORT(T)+1530./T  
IF (T .LT. 5000.) GO TO 10  
CPO = CPO\*5.0E-05\*(T-4000.)  
10 CPAIR = FMO\*CPO/WMO+FMN\*CPN/WMN  
RETURN  
END

END OF COMPILATION: NO DIAGNOSTICS.

QHD6,P \*\*\*\*\* CPF/CFFC43 \*\*\*\*\*

\*\*\*\*\* CPF/CFFC43 \*\*\*\*\*

QFOR,S ME\*NASAS.CPF/CFFC43,ME\*NASAS5.CPF/CFFC43  
FOR S9A-07/13/72-20:54:41 (0,)

FUNCTION CPF ENTRY POINT 000014

STORAGE USED: CODE(1) 000016; DATA(0) 000011; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3S

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000000 CPF 0000 000004 INJPS 0000 R 000001 X1 0000 R 000002 X2

00101	1*	C	FUNCTION CPF(RHO*T)
00101	2*	C	
00101	3*	C	THIS FUNCTION SUBPROGRAM COMPUTES :
00101	4*	C	SPECIFIC HEAT AT CONSTANT PRESSURE AS A FUNCTION OF
00101	5*	C	DENSITY (SLUG/CU.FT) AND TEMPERATURE (R) OF FC-43
00101	6*	C	UNITS BTU/(SLUG.R)
00101	7*	C	
00103	8*		DATA X1,X2 /-0.020092;5.4054E-04/
00106	9*		CPF = (X1+X2*T)*32.174
00107	10*		RETURN
00110	11*		END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* CPF/CFFC75 \*\*\*\*\*

\*\*\*\*\* CPF/CF75 \*\*\*\*\*  
FOR S ME\*NASAS.CPF/CF75\*ME\*NASAS.CPF/CF75  
FOR S9A-07/13/72-20:54:43 (0,)

FUNCTION CPF ENTRY POINT 00014

STORAGE USED: CODE(1) 000016; DATA(0) 000011; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000000 CPF 0000 000004 INJPS 0000 R 000001 X1 0000 R 000002 X2

00101	1*	FUNCTION CPF(RHO,T)
00101	2*	C
00101	3*	C THIS FUNCTION SUBPROGRAM COMPUTES :
00101	4*	C SPECIFIC HEAT AT CONSTANT PRESSURE AS A FUNCTION OF
00101	5*	C DENSITY (SLUG/CU.FT) AND TEMPERATURE (R) OF FC-75
00101	6*	C UNITS BTU/(SLUG.R)
00101	7*	C
00103	8*	DATA X1,X2 /0.115082,2.4333E-04/
00106	9*	CPF = (X1+X2*T)*32.174
00107	10*	RETURN
00110	11*	END

END OF COMPILATION: NO DIAGNOSTICS.

ENDG:P \*\*\*\*\* CPF/CFHE \*\*\*\*\*

DFOR,S ME\*NASAS.CPF/CFHE,ME\*NASAS.CPF/CFHE  
FOR S9A-07/13/72-20:54:45 (0,)

FUNCTION CPF ENTRY POINT 000057

STORAGE USED: CODE(1) 000071; DATA(0) 000031; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 BETA  
0004 CAPPA  
0005 NERR3S

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000005 A 0000 R 000010 ALPHA 0000 R 000002 A1 0003 R 000000 BETA 0000 R 000003 BI  
0000 R 000004 C 0004 R 000000 CAPPA 0000 R 000000 CPF 0000 R 000011 CV 0000 000024 INJPS  
0000 R 000001 R 0000 R 000006 RHOX 0000 R 000007 TX 0000 R 000012 XI

00101 1\* FUNCTION CPF (RHO,T)

00101 2\* C  
00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
00101 4\* C SPECIFIC HEAT AT CONSTANT PRESSURE AS A FUNCTION OF  
00101 5\* C DENSITY (SLUG/CU.FT) AND TEMPERATURE (R) OF HELIUM  
00101 6\* C UNITS BTU/(SLUG.R)  
00101 7\* C  
00103 8\* DATA R,A1,B1,C,A /2077.02,136.9595,3.5002295E-03,10.000558,1.49610  
00103 9\* 13E-02/  
00111 10\* RHOX = RHO\*515.4275  
00112 11\* TX = T/1.8  
00113 12\* ALPHA = C\*RHOX/(TX\*\*3)  
00114 13\* CV = R\*(1.5+6.0\*ALPHA\*(1.0+RHOX\*B1\*0.5))  
00115 14\* X1 = BETA(RHO,T)\*1.8  
00116 15\* CPF = (CV\*(TX\*X1\*X1\*47.872)/(RHOX\*CAPPA(RHO,T)))\*76.8624E-04  
00117 16\* RETURN  
00120 17\* END

END OF COMPILATION: NO DIAGNOSTICS.

ENDG,P \*\*\*\*\* CPF/CFNAK \*\*\*\*\*



9FOR'S MERNASAS.CPF/CFNAK,ME\*NASAS5.CPF/CFNAK  
FOR S9A-07/13/72-20:54:47 (0.)

FUNCTION CPF ENTRY POINT 000071

STORAGE USED: CODE(1) 000104; DATA(0) 000032; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 CAPPA  
0004 PF  
0005 ALOG  
0006 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0003 R	000000	CAPPA	0000 R	000014	CP	0000 R	000000	CPF	0000 R	000010	C1
0000 R	000007	UT	0000 R	000023	INJPS	0004 R	000000	PF	0000 R	000012	P0
0000 R	000013	THETA	0000 R	000015	TK	0000 R	000006	T0	0000 R	000001	X1
0000 R	000003	X3	0000 R	000004	X4	0000 R	000005	X5	0000 R	000002	X2

FUNCTION CPF(RHO,T)

00101 1\* C  
00101 2\* C  
00101 3\* C  
00101 4\* C  
00101 5\* C  
00101 6\* C  
00101 7\* C  
00101 8\* C  
00103 9\* C  
00116 10\* C  
00117 11\* C  
00120 12\* C  
00121 13\* C  
00122 14\* C  
00122 15\* C  
00123 16\* C  
00124 17\* C

THIS FUNCTION SUBPROGRAM COMPUTES:  
SPECIFIC HEAT AT CONSTANT PRESSURE AS A FUNCTION OF  
DENSITY (SLUG/CU.FT) AND TEMPERATURE (R) OF NAK 78.6  
UNITS BTU/(SLUG.R)

DATA X1,X2,X3,X4,X5 /0.2255,-0.016292,0.005396,-0.000758,0.000054/  
1,T0,DT /659.67,300.0/\*C1,C2 /58.773064,-0.008433/P0 /2116.224/  
THETA = (T-T0)/DT  
CP = (((X5\*THETA+X4)\*THETA+X3)\*THETA+X2)\*THETA+X1  
TK = CAPPA(RHO,T)  
P1 = PF(RHO,T)-P0  
CPF = (CP-(2.0\*T\*C2\*C2\*ALOG(1.0+TK\*P1)))/(TK\*778.26\*(C1+C2\*T)\*\*3)  
1)\*32.174  
RETURN  
END

END OF COMPILATION: NO DIAGNOSTICS.

9HDG/P \*\*\*\*\* CPF/CFSIL \*\*\*\*\*

\*\*\*\*\* CPF/CFSIL \*\*\*\*\*

QFOR: S ME\*NASAS.CPF/CFSIL, ME\*NASAS.CPF/CFSIL  
 FOR S9A-07/13/72-20:54:50 (0,)

FUNCTION CPF ENTRY POINT 000350

STORAGE USED: CODE(1) 000372; DATA(0) 000133; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 PF  
 0004 POLY  
 0005 NWDUS  
 0006 N102\$  
 0007 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000	000055	1F	0000 R 000040 A	0000 R 000007 A1	0000 R 000041 A11	0000 R 000010 A2
0000	R 000042	A22	0000 R 000011 A3	0000 R 000012 A4	0000 R 000013 A5	0000 R 000043 B
0000	R 000014	B1	0000 R 000044 B11	0000 R 000015 B2	0000 R 000045 B22	0000 R 000016 B3
0000	R 000017	B4	0000 R 000020 B5	0000 R 000046 C	0000 R 000037 CP	0000 R 000000 CPF
0000	R 000021	C1	0000 R 000047 C11	0000 R 000022 C2	0000 R 000025 DT	0000 R 000033 DT2
0000	R 000050	E1	0000 R 000051 E2	0000 R 000052 E3	0000 R 000054 FP	0000 R 000113 INJPs
0000	R 000053	P	0003 R 000000 PF	0004 R 000000 POLY	0000 R 000034 THETA	0000 R 000035 THETA1
0000	R 000036	THETA2	0000 R 000023 T0	0000 R 000024 T01	0000 R 000032 T02	0000 R 000026 X1
0000	R 000027	X2	0000 R 000030 X3	0000 R 000031 X4	0000 R 000001 Z	

FUNCTION CPF (RHO,T)

00101	1*	C	
00101	2*	C	
00101	3*	C	
00101	4*	C	
00101	5*	C	
00101	6*	C	
00101	7*	C	
00101	8*	C	
00101	9*	C	
00103	10*		
00104	11*		
00104	12*		
00104	13*		
00132	14*		
00133	15*		
00134	16*		
00135	17*		
00136	18*		
00137	19*		
00140	20*		
00141	21*		
00142	22*		
00143	23*		

THIS FUNCTION SUBPROGRAM COMPUTES :  
 SPECIFIC HEAT AS A FUNCTION OF DENSITY (SLUG/CU.FT) AND  
 TEMPERATURE (R) OF DOW CORNING 200 SILICON OIL (1 CS)  
 UNITS BTU/(SLUG.R)  
 TEMPERATURE .GE. 359.67 AND .LE. 859.67

DIMENSION Z(6)  
 DATA A1,A2,A3,A4,A5 /12.35,2.98333,1.1,-0.48333,0.1/,B1,B2,B3,B4,B  
 15 /-1.5,-0.01333,-1.18,0.57333,-0.1/,C1,C2 /0.7767,-0.0288/,T0,T01  
 2,DT /559.67,609.67,50.0/,X1,X2,X3,X4 /0.46,0.00471,0.00141,0.00004  
 33/,T02,DT2 /539.67,80.0/  
 THETA = (T-T01)/DT  
 THETA1 = (T-T02)/DT2  
 THETA2 = (T-T02)/DT2  
 CP = ((X4\*THETA2+X3)\*THETA2+X2)\*THETA2+X1  
 A = (((A5\*THETA+A4)\*THETA+A3)\*THETA+A2)\*THETA+A1)\*1.0E-06  
 A11 = (((4.0\*A5\*THETA+3.0\*A4)\*THETA+2.0\*A3)\*THETA+A2)\*1.0E-06/DT  
 A22 = (((12.0\*A5\*THETA+6.0\*A4)\*THETA+2.0\*A3)\*THETA+A2)\*1.0E-06/DT  
 B = (((B5\*THETA+B4)\*THETA+B3)\*THETA+B2)\*THETA+B1)\*1.0E-09  
 B11 = (((4.0\*B5\*THETA+3.0\*B4)\*THETA+2.0\*B3)\*THETA+B2)\*1.0E-09/DT  
 B22 = (((12.0\*B5\*THETA+6.0\*B4)\*THETA+2.0\*B3)\*THETA+B2)\*1.0E-09/DT

\*\*\*\*\* CPF/CFSIL \*\*\*\*\*

```

00144 C = C1+C2*THETA1
00145 C11 = C2/DT
00146 E1 = A11*A11-B22/2.0+811*C11/C
00147 E2 = 2.0*A11*C11/C-A22
00150 E3 = A*A-B
00151 P = PF(RHO,T)/144.0-14.696
00152 Z(1) = 0.0
00153 Z(2) = 2.0*(C11/C)**2
00154 Z(3) = (E2-A*Z(2))/2.0
00155 Z(4) = (E1-A*E2+E3*Z(2)/2.0)/3.0
00156 Z(5) = (A11*B11-A*E1+E3*E2/2.0+A*B*Z(2)/2.0)/4.0
00157 Z(6) = (B11*B11/2.0-2.0*A*A11*B11+E1*E3+A*B*E2*B*Z(2)/2.0)/10.0
00160 FP = POLY(6,Z,P)
00161 CPF = (CP-T*FP/(C*337.37))*32.174
00162 IF (T.GE.360.67.AND.T.LE.860.67) RETURN
00164 WRITE (6,1) T
00167 1 FORMAT (1H0.54HSPECIFIC HEAT OF SILICON OIL, TEMP. OUT OF RANGE, T
00167 1 = ,F8.3,/)
00170 RETURN
00171 END

```

END OF COMPILATION: NO DIAGNOSTICS.

\*\*\*\*\* CPFN/FNAL \*\*\*\*\*

QFOR.S ME\*NASAS.CPFN/FNAL,ME\*NASAS.CPFN/FNAL  
 FOR 59A-07/13/72-20:54:54 (0,)

FUNCTION CPFN ENTRY POINT 000056

STORAGE USED: CODE(1) 000062; DATA(0) 000040; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 N4DJ\$  
 0004 NI02\$  
 0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000012 1F 0000 R 000011 CP 0000 R 000000 CPFN 0000 R 000007 DT 0000 000032 INJPS  
 0000 R 000010 THETA 0000 R 000006 T0 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3  
 0000 R 000004 X4 0000 R 000005 X5

00101 1\* FUNCTION CPFN(T)

00101 2\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
 00101 3\* C SPECIFIC HEAT AS A FUNCTION OF TEMPERATURE (R) OF ALUMINUM 7075  
 00101 4\* C UNITS BTU/(SLUG.R)  
 00101 5\* C TEMP. GE. 300.0.AND.LE. 1200.0 R  
 00101 6\* C  
 00101 7\* C  
 00103 8\* DATA X1,X2,X3,X4,X5 /0.182+0.03616,-0.011417+0.00233,-0.000083/.T0  
 00103 9\* 1,DT /400.0,200.0/  
 00113 10\* THETA = (T-T0)/DT  
 00114 11\* CP = (((X5\*THETA+X4)\*THETA+X3)\*THETA+X2)\*THETA+X1  
 00115 12\* CPFN = CP\*32.174  
 00116 13\* IF (T-GE.300.0.AND.T.LE.1200.0) RETURN  
 00120 14\* WRITE (6,1) T  
 00123 15\* 1 FORMAT (1H0.51HSPECIFIC HEAT OF ALUMINUM, TEMP. OUT OF RANGE, T =  
 00124 16\* 1,F8.3,/) RETURN  
 00125 17\* END  
 00125 18\* END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG.P \*\*\*\*\* CPFN/FNBR \*\*\*\*\*

\*\*\*\*\* CPFN/FNBR \*\*\*\*\*

9FOR 5 ME\*NASAS.CPFN/FNBR\*ME\*NASAS.CPFN/FNBR  
FOR 59A-07/13/72-20:54:57 (0.)

FUNCTION CPFN ENTRY POINT 000054

STORAGE USED: CODE(1) 000060; DATA(0) 000037; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDJ\$  
0004 NIO2\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000011 1F 0000 R 000010 CP 0000 R 000000 CPFN 0000 R 000006 DT 0000 000031 INJPS  
0000 R 000007 THETA 0000 R 000005 T0 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3  
0000 R 000004 Xi

00101 1\* FUNCTION CPFN(T)

00101 2\* C  
00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
00101 4\* C SPECIFIC HEAT AS A FUNCTION OF TEMPERATURE (R) OF BERYLLIUM WITH  
00101 5\* C 0.84-1.68 BEQ UNITS BTU/(SLUG.R)  
00101 6\* C TEMP. GE. 400.0 AND LE. 1700.0 R  
00101 7\* C  
00101 8\* C  
00101 9\* C  
00103 10\* C  
00112 11\* CP = ((X4\*THETA+X3)\*THETA+X2)\*THETA+X1  
00113 12\* CPFN = CP\*32.174  
00114 13\* IF (T.GE.400.0.AND.T.LE.1700.0) RETURN  
00115 14\* WRITE (6,1) T  
00117 15\* 1.FORMAT (1H0,51HSPECIFIC HEAT OF BERYLLIUM= TEMP. OUT OF RANGE, T =  
00122 16\* 1.FB.3./)  
00122 17\* RETURN  
00123 18\* END  
00124

END OF COMPILE: NO DIAGNOSTICS.

DHDG.P \*\*\*\*\* CPFN/FNBU \*\*\*\*\*

\*\*\*\*\* CPFN/FNCU \*\*\*\*\*

FOR S ME\*NASAS.CPFN/FNCU,ME\*NASAS.CPFN/FNCU  
 FOR S9A-07/13/72-20:54:59 (0,)

FUNCTION CPFN ENTRY POINT 000045

STORAGE USED: CODE(1) 000051; DATA(0) 000031; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDUS  
 0004 NIO23  
 0005 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000004 1F 0000 R 000003 CP 0000 R 000000 CPFN 0000 000023 INJPS 0000 R 000001 X1  
 0000 R 000002 X2

```

00101 1*          FUNCTION CPFN(T)
00101 2*          C
00101 3*          C      THIS FUNCTION SUBPROGRAM COMPUTES :
00101 4*          C      SPECIFIC HEAT AS A FUNCTION OF TEMPERATURE (R) OF COPPER
00101 5*          C      UNITS BTU/(SLUG.R)
00101 6*          C      TEMP. 6E. 400.0 AND LE. 1800.0 R
00101 7*          C
00103 8*          DATA X1,X2 /0.08375,1.375E-05/
00106 9*          CP = X2*T+X1
00107 10*         CPFN = CP*32.174
00110 11*         IF (T.GE.400.0.AND.T.LE.1800.0) RETURN
00112 12*         WRITE (6,1) T
00115 13*         1 FORMAT (1H0,49HSPECIFIC HEAT OF COPPER, TEMP. OUT OF RANGE, T = ,
00115 14*         1F8.3,/)
00116 15*         RETURN
00117 16*         END
  
```

END OF COMPILATION: NO DIAGNOSTICS.

ENDG:P \*\*\*\*\* CPMP/NPAL \*\*\*\*\*

\*\*\*\*\* CPMP/MPAL \*\*\*\*\*

FOR S ME\*NASAS.CPMP/MPAL,ME\*NASAS.CPMP/MPAL  
FOR S9A-07/13/72-20:55:01 (0.)

FUNCTION CPMP ENTRY POINT 000056

STORAGE USED: CODE(1) 000062; DATA(0) 000040; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDJ\$  
0004 NIO2\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000012 IF 0000 R 000011 CP 0000 R 000000 CPMP 0000 000032 INJPS  
0000 R 000010 THETA 0000 R 000006 T0 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3

R 0000 00000 4 X 4 0000 00000 X 5 5

00101 1\* FUNCTION CPMP(T)

00101 2\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
00101 3\* C SPECIFIC HEAT AS A FUNCTION OF TEMPERATURE (R) OF ALUMINUM 7075  
00101 4\* C UNITS BTU/(SLUG\*R)  
00101 5\* C TEMP. GE. 300.0.AND.LE. 1200.0 R  
00101 6\* C  
00101 7\* C  
00101 8\* DATA X1,X2,X3,X4,X5 /0.182\*0.03616,-0.011417\*0.00233,-0.000083/,T0  
00103 9\* 1,DT /400.0,200.0/  
00113 10\* THETA = (T-T0)/DT  
00114 11\* CP = ((X5\*THETA+X4)\*THETA+X3)\*THETA+X2)\*THETA+X1  
00115 12\* CPMP = CP\*32.174  
00116 13\* IF (T-GE.300.0.AND.T.LE.1200.0) RETURN  
00120 14\* WRITE (6,1) T  
00123 15\* 1,FORMAT (1H0,51HSPECIFIC HEAT OF ALUMINUM, TEMP. OUT OF RANGE, T =  
00123 16\* 1,F8.3,/  
00124 17\* RETURN  
00125 18\* END

END OF COMPILATION: NO DIAGNOSTICS.

QHDP \*\*\*\*\* CPMP/MPBR \*\*\*\*\*

QFOR'S ME\*NASAS.CPMP/MPBR, ME\*NASAS5.CPMP/MPBR  
FOR S9A-071372-20:55:03 (0\*)

FUNCTION CPMP ENTRY POINT 000054

STORAGE USED: CODE(1) 000060; DATA(0) 000037; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDUS  
0004 NI02\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000011 IF 0000 R 000010 CP 0000 R 000006 DT 0000 000031 INJPS  
0000 R 000007 THETA 0000 R 000005 T0 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3  
0000 R 000004 X4

00101 1\* FUNCTION CPMP(T)  
00101 2\* C  
00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES:  
00101 4\* C SPECIFIC HEAT AS A FUNCTION OF TEMPERATURE (R) OF BERYLIUM WITH  
00101 5\* C 0.84-1.68 BE0  
00101 6\* C UNITS BTU/(SLUG.R)  
00101 7\* C TEMP. GE. 400.0 AND LE. 1700.0 R  
00101 8\* C  
00103 9\* DATA X1,X2,X3,X4 /0.536,0.05667,-0.0085,0.00083/,T0,DT /800.,200./  
00112 10\* THETA = (T-T0)/DT  
00113 11\* CP = ((X4\*THETA+X3)\*THETA+X2)\*THETA+X1  
00114 12\* CPMP = CP\*32.174  
00115 13\* IF (T-GE.400.0.AND.T.LE.1700.0) RETURN  
00117 14\* WRITE (6,1) T  
00122 15\* 1 FORMAT (1H0,5HSPECIFIC HEAT OF BERYLIUM= TEMP. OUT OF RANGE, T =  
00122 16\* 1,F8.3,/) RETURN  
00123 17\* END  
00124 18\*

END OF COMPILATION: NO DIAGNOSTICS.

ENDG,P \*\*\*\*\* CPMP/MPCU \*\*\*\*\*



QFOR,S ME\*NASAS5.CPMP/MPCU,ME\*NASAS5.CPMP/MPCU  
FOR S9A-07/13/72-20:55:06 (0,)

FUNCTION CPMP ENTRY POINT 000045

STORAGE USED: CODE(1) 000051; DATA(0) 000031; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDJ\$  
0004 NIO2\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000004 1F 0000 R 000003 CP 0000 R 000000 CPMP 0000 000023 INJPS 0000 R 000001 X1  
0000 R 000002 X2

```

00101 1*          FUNCTION CPMP(T)
00101 2*          C
00101 3*          C      THIS FUNCTION SUBPROGRAM COMPUTES :
00101 4*          C      SPECIFIC HEAT AS A FUNCTION OF TEMPERATURE (R) OF COPPER
00101 5*          C      UNITS BTU/(SLUG-R)
00101 6*          C      TEMP, GE, 400.0 AND LE, 1800.0 R
00101 7*          C
00103 8*          C      DATA X1,X2 /0.08375,1.375E-05/
00106 9*          C      CP = X2*T+X1
00107 10*         C      CPMP = CP*32.174
00110 11*         C      IF (T.GE.400.0.AND.T.LE.1800.0) RETURN
00112 12*         C      WRITE (6,1) T
00115 13*         C      1 FORMAT (1H0,49HSPECIFIC HEAT OF COPPER, TEMP. OUT OF RANGE, T = ,
00115 14*         C      1F8.3,/)
00116 15*         C      RETURN
00117 16*         C      END

```

END OF COMPILATION: NO DIAGNOSTICS.

QHDG:P \*\*\*\*\* CPT8/TBAL \*\*\*\*\*

QFOR'S ME\*NASAS5.CPTB/TBAL,ME\*NASAS5.CPTB/TBAL  
FOR S9A-07/13/72-20:55:08 (0,)

FUNCTION CPTB ENTRY POINT 000056

STORAGE USED: CODE(1) 000062; DATA(0) 000040; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDJ\$  
0004 NI02\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000012 1F 0000 R 000011 CP 0000 R 000000 CPTB 0000 R 000007 DT 0000 000032 INJPS  
0000 R 000010 THETA 0000 R 000006 T0 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3  
0000 R 000004 X4 0000 R 000005 X5

00101 1\* FUNCTION CPTB(T)

00101 2\* C  
00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
00101 4\* C SPECIFIC HEAT AS A FUNCTION OF TEMPERATURE (R) OF ALUMINUM 7075  
00101 5\* C UNITS BTU/(SLUG.R)  
00101 6\* C TEMP. GE. 300.0.AND.LE. 1200.0 R  
00101 7\* C  
00101 8\* DATA X1,X2,X3,X4,X5 /0.182,0.03616,-0.0114,17,0.00233,-0.000083/,T0  
00103 9\* 1,DT /400.0,200.0/  
00113 10\* THETA = (T-T0)/DT  
00114 11\* CP = ((X5\*THETA+X4)\*THETA+X3)\*THETA+X2)\*THETA+X1  
00115 12\* CPTB = CP\*32.174  
00116 13\* IF (T,GE,300.0.AND,T.LE,1200.0) RETURN  
00120 14\* WRITE (6,1) T  
00123 15\* 1 FORMAT (1H0,51HSPECIFIC HEAT OF ALUMINUM, TEMP. OUT OF RANGE, T =  
00123 16\* 1,F6.3,/) RETURN  
00124 17\* END  
00125 18\* END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG\*P \*\*\*\*\* CPTB/TBBR \*\*\*\*\*

9FOR:5 ME\*NASAS.CPTB/TBBR,ME\*NASAS.CPTB/TBBR  
FOR S9A-07/13/72-20:55:10 (0.)

FUNCTION CPTB ENTRY POINT 000054

STORAGE USED: CODE(1) 000060; DATA(0) 000037; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDUS  
0004 NIO2\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000011 IF 0000 R 000010 CP 0000 R 000000 CPTB 0000 R 000006 DT 0000 000031 INJPS  
0000 R 000007 THETA 0000 R 000005 T0 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3  
0000 R 000004 X4

00101 1\* FUNCTION CPTB(T)

00101 C  
00101 C THIS FUNCTION SUBPROGRAM COMPUTES :  
00101 C SPECIFIC HEAT AS A FUNCTION OF TEMPERATURE (R) OF BERYLIUM WITH  
00101 C 0.84--1.68 BEO UNITS BTU/(SLUG.R)  
00101 C TEMP. GE. 400.0 AND LE. 1700.0 R  
00101 C  
00103 C DATA X1,X2,X3,X4 /0.536,0.05667,-0.0085,0.00083/,T0,DT /800.,200./  
00112 C THETA = (T-T0)/DT  
00113 C CP = ((X4\*THETA+X3)\*THETA+X2)\*THETA+X1  
00114 C CPTB = CP+32.174  
00115 C IF (T.GE.400.0.AND.T.LE.1700.0) RETURN  
00117 C WRITE (6,1) T  
00122 C 1 FORMAT (I10,5IHSPECIFIC HEAT OF BERYLIUM TEMP. OUT OF RANGE, T =  
00122 C 1,FB,3,/) RETURN  
00123 C 17\*  
00124 C 18\* END

END OF COMPILATION: NO DIAGNOSTICS.

9HDG:P \*\*\*\*\* CPTB/TBCU \*\*\*\*\*

\*\*\*\*\* CPTB/TBCU \*\*\*\*\*

QFOR,S ME\*NASAS.CPTB/TBCU,ME\*NASAS.CPTB/TBCU  
 FOR S9A-07/13/72-20:55:13 (0,)

FUNCTION CPTB ENTRY POINT 000045

STORAGE USED: CODE(1) 000051; DATA(0) 000031; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDJ\$  
 0004 NI02\$  
 0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000004 1F 0000 R 000003 CP 0000 R 000000 CPTB 0000 000023 INJPS 0000 R 000001 X1  
 0000 R 000002 X2

```

00101 1*          FUNCTION CPTB(T)
00101 2*          C
00101 3*          C      THIS FUNCTION SUBPROGRAM COMPUTES :
00101 4*          C      SPECIFIC HEAT AS A FUNCTION OF TEMPERATURE (R) OF COPPER
00101 5*          C      UNITS BTU/(SLUG.R)
00101 6*          C      TEMP. 6E. 400.0 AND LE. 1800.0 R
00101 7*          C
00103 8*          DATA X1,X2 /0.08375,1.375E-05/
00106 9*          CP = X2*I+X1
00107 10*         CPTB = CP*32.174
00110 11*         IF (T.GE.400.0.AND.T.LE.1800.0) RETURN
00112 12*         WRITE (6,1) T
00115 13*         1 FORMAT (1H0,49HSPECIFIC HEAT OF COPPER, TEMP. OUT OF RANGE, T = ,
00116 14*         IF8.3,/)
00116 15*         RETURN
00117 16*         END
  
```

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* ODX \*\*\*\*\*

\*\*\*\*\* DDX \*\*\*\*\*

OFOR,S ME\*NASAS.DDX,ME\*NASAS.DDX  
FOR S9A-07/13/72-20:55:15 (0,)

SUBROUTINE DDX ENTRY POINT 000127

STORAGE USED: CODE(1) 000153; DATA(0) 000046; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NW00S  
0004 NI02S  
0005 NERR3S

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000061 1129 0001 000102 2L 0000 000003 3F 0000 R 000000 DX1 0000 I 000002 I  
0000 000016 INJPS 0000 I 000001 N1

00101	1*		SUBROUTINE DDX (Y,DY,DX,N)	1
00101	2*	C		
00101	3*	C	THIS SUBROUTINE COMPUTES :	
00101	4*	C	THE FIRST DERIVATIVE OF THE EQUALLY-SPACED ARRAY Y(N)	
00101	5*	C		
00103	6*		DIMENSION Y(N),DY(N)	2
00104	7*		IF (N.LT.3) GO TO 2	
00106	8*		DX1 = 2.0*DX	5
00107	9*		DY(1) = (-3.0*Y(1)+4.0*Y(2)-Y(3))/DX1	6
00110	10*		N1 = N-1	7
00111	11*		DO 1 I=2,N1	8
00114	12*		1 DY(I) = (Y(I+1)-Y(I-1))/DX1	10
00116	13*		DY(N) = (Y(N-2)-4.0*Y(N-1)+3.0*Y(N))/DX1	11
00117	14*		RETURN	12
00120	15*		2 WRITE (6,3)	13
00122	16*		3 FORMAT (25HDDX NOT ENOUGH NODAL PTS.,/)	14
00123	17*		RETURN	
00124	18*		END	

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* DEFINIT \*\*\*\*\*

\*\*\*\*\* DEFINIT \*\*\*\*\*  
 QFOR,S ME\*NASAS5,DEFINT,ME\*NASAS5,DEFINT  
 FOR S9A-07/13/72-20:53:18 (0,)

FUNCTION DEFINIT ENTRY POINT 000203

STORAGE USED: CODE(1) 000225; DATA(0) 000046; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3S

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000166	10L	0001	000117	131G	0001	000126	136G	0001	000032	2L	0001	000054	4L
0000	R	000000	DEFINT	0000	000024	INJP\$	0000	I	000011	K	0000	I	000006	NC
0000	I	000010	NEVE	0000	I	000007	NODD	0000	I	000003	N1	0000	I	000004
0000	R	000002	SODD									0000	R	000001
														SEVE

FUNCTION DEFINIT(Y,DX,N)

1\* C  
 2\* C  
 3\* C  
 4\* C  
 5\* C  
 6\* C  
 7\* C  
 8\* C  
 9\* C  
 10\* C  
 11\* C  
 12\* C  
 13\* C  
 14\* C  
 15\* C  
 16\* C  
 17\* C  
 18\* C  
 19\* C  
 20\* C  
 21\* C  
 22\* C  
 23\* C  
 24\* C  
 25\* C  
 26\* C  
 27\* C  
 28\* C  
 29\* C  
 30\* C  
 31\* C

T HIS FUNCTION SUBPROGRAM COMPUTES :  
 THE DEFINITE INTEGRAL BY SIMPSON'S RULE OF THE EQUALLY-SPACED  
 ARRAY Y(N)  
 DIMENSION Y(N)  
 IF(N-3)10,1,2  
 1 DEFINT = (DX/3.)\*(Y(1)+4.\*Y(2)+Y(3))  
 RETURN  
 2 IF(N-4)10,3,4  
 3 DEFINT = (3.\*DX/8.)\*(Y(1)+3.\*Y(2)+Y(3))+Y(4))  
 RETURN  
 4 SEVE = 0.  
 SODD = 0.  
 N1 = N/2  
 N2 = 2\*N1  
 M = N  
 NC = M-N2  
 IF(NC.EQ.0) M=N-1  
 NODD = M-1  
 NEVE = M-2  
 DO 7 K=2,NODD,2  
 7 SODD = SODD+Y(K)  
 DO 8 K=3,NEVE,2  
 8 SEVE = SEVE+Y(K)  
 DEFINT=(DX/3.)\*(Y(1)+Y(M)+4.\*SODD+2.\*SEVE)  
 IF(NC)10,9,10  
 9 DEFINT = DEFINT+(DX/24.)\*(19.\*Y(N)+19.\*Y(N-1)-5.\*Y(N-2)+Y(N-3))  
 10 RETURN  
 END

\*\*\*\*\* DEFINT \*\*\*\*\*

END OF COMPILATION: NO DIAGNOSTICS.

QHDG\*P \*\*\*\*\* DEFINT \*\*\*\*\*

\*\*\*\*\* DEFNT \*\*\*\*\*

DEFOR,S ME\*NASAS,DEFNT,ME\*NASAS,DEFNT  
FOR S9A-07/13/72-20:55:22 (0,)

FUNCTION DEFNT ENTRY POINT 000046

STORAGE USED: CODE(1) 000060; DATA(0) 000023; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000020 1076 0000 R 000000 DEFNT 0000 I 000003 I 0000 000007 INJPS 0000 I 000001 L1  
0000 R 000002 Z

00101	1*		FUNCTION DEFNT(Y,DX,N)
00101	2*	C	
00101	3*	C	THIS SUBROUTINE COMPUTES :
00101	4*	C	THE DEFINITE INTEGRAL BY TRAPEZOIDAL RULE OF THE EQUALLY-SPACED
00101	5*	C	ARRAY Y(N)
00101	6*	C	
00103	7*		DIMENSION Y(N)
00104	8*		L1 = N-1
00105	9*		Z = 0.0
00106	10*		DO 5 I= 2,L1
00111	11*		5 Z = Z+Y(I)
00113	12*		DEFNT = DX*(Y(1)+2.0*Z+Y(N))/2.0
00114	13*		RETURN
00115	14*		END

END OF COMPILATION: NO DIAGNOSTICS.

0H06:P \*\*\*\*\* DERIVL \*\*\*\*\*



\*\*\*\*\* DERIVL \*\*\*\*\*

DFOR,S ME\*NASAS.DERIVL,ME\*NASAS.DERIVL  
FOR S9A-07/13/72-20:55:25 (07)

SUBROUTINE DERIVL ENTRY POINT 000245

STORAGE USED: CODE(1) 000257; DATA(0) 000053; BLANK COMMON(2) 001115

COMMON BLOCKS:

0003 FLCMP 000024

EXTERNAL REFERENCES (BLOCK, NAME)

0004 RHOF  
0005 CAPPA  
0006 BETA  
0007 CPF  
0010 NERR2\$  
0011 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000177	L9L	0001	000132	5L	0002	000562	AUX2	0000	R	000013	A13
0000	R	000017	A21	0000	R	000020	A23	0000	R	000014	A31	0000
0006	R	000000	BETA	0000	R	000021	B1	0000	R	000022	B2	0002
0002	001107	CJH	0002	000574	CONFN	0002	000740	CONMP	0007	R	000000	CAPPA
0000	R	000023	C1	0000	R	000024	C2	0000	R	000025	C3	0002
0003	R	000000	UTHEA	0002	001110	DXI	0002	001053	DXIMP	0002	001072	DELTA
0002	001103	FENTH	0002	001045	FFO	0002	001101	FLUX	0002	R	001052	DXITB
0002	001046	FPF	0002	000012	FP	0002	001042	FPR	0002	R	001040	FNU
0002	001111	FRO	0002	001036	FRE	0002	001104	FREJ	0002	R	001037	FR
0002	R	000036	FT	0002	000024	FW	0002	001113	FXHW	0002	001043	FRL
0000	R	000010	G3ETA	0000	R	000011	GCP	0000	R	001105	HFN	IFLOW
0000	000030	I4VPS	0002	001024	LC	0002	001032	LIM	0002	001031	LTMWRT	0002
0002	001012	LL2	0002	001013	LL3	0002	001014	LL4	0002	001015	LL5	0002
0002	001017	LL7	0002	001006	LMP	0002	001004	LT	0002	001005	LTR	0002
0002	001010	LT84Z	0002	001030	MM1	0002	001033	MOD	0002	001025	45TOTR	0002
0002	001034	NCCZ	0002	001035	NCONV	0002	I	001002	NCTL	0002	001026	NEGUS
0002	001027	N41	0002	001001	NRMP	0002	001020	NRMP1	0002	001022	NRMP2	0002
0002	001021	NR1B1	0002	000776	NX	0000	R	000000	P	0002	001063	PHIF
0002	001071	PI	0000	R	000003	PP	0002	R	001065	PO	0004	R
0002	000524	CRMP	0002	001074	ROTWRT	0000	R	000006	RHO	0002	001102	GHOF
0002	R	001067	PH00	0002	001073	RLIMIT	0000	R	000005	RRHO	0002	001114
0002	001032	TEMP	0002	001077	TI	0002	001076	TINTL	0002	000276	TMP	0000
0000	R	000004	TT	0002	R	000050	TW	0002	R	000001	W	0002
0002	000536	XIFN	0002	000005	XIMP	0002	000000	XITB	0002	001106	XL	0002
0002	001057	XX10	0002	001060	XX11	0002	001061	XX12	0002	001055	XX3	0002
0002	000550	ZETA										

00101 1\* SUBROUTINE DERIVL (Y,DY,X)

\*\*\*\*\* DERIVL \*\*\*\*\*

```

00101 2* C
00101 3* C THIS SUBROUTINE COMPUTES :
00101 4* C SPATIAL DERIVATIVES DY OF Y
00101 5* C
00103 6* COMMON XITB(5),XIMP(5),FP(10),FW(10),FT(10),TW(10,5),TEMP(10,10),
00103 7* TMP(10,5),GRFN(10,10),GRMP(10),XIFN(10),ZETA(10),AUX2(10),
00103 8* CONFN(10,10),CONMP(10),COB(10),DOB(10)
00104 9* COMMON NX,M2,NRTB,NRMP,NCTL,IFLOW,LT,LTB,LMP,LIBMZ,LTB2MZ,LL1,
00104 10* LL2,LL3,LL4,LL5,LL6,LL7,NRMP1,NRTB1,NRMP2,NEQUS,LC,MSTOTR,
00104 11* NCTM,NMI,MM1,IMWRT,LIM,MOD,NCCZ,NCONV
00105 12* COMMON FRE,FR,FNU,FMACH,FPR,FRL,FRM,FFO,FOF,FZ,FRAD,DZ,DXITB,DXIMP
00105 13* ,XRE,XX3,XX4,XX10,XX11,XX12,PHIM,PHIF,TO,PO,W0,RH00,CP0,PI
00105 14* ,DELTA,RLIMIT,RDYWT,TREF,TINTL,TI,RTEND,FLUX,QREF,FENTH,
00105 15* FREJ,HFN,XL,COH,DXI,FRD,FXOH,FXHM,RHOFN
00106 16* COMMON/FLC4P/DTHETA(20)
00106 17* C
00107 18* DIMENSION Y(3),DY(3)
00107 19* C
00110 20* P = Y(1)
00111 21* W = Y(2)
00112 22* T = Y(3)
00113 23* IF (IFLOW .EQ. 2) T = FT(NCTL)
00113 24* C
00115 25* PP = P0*P
00116 26* TT = T0*T
00117 27* RRHO = RHOF(PP,TT)
00120 28* RHO = RRHO/RH00
00121 29* GKAPPA = CAPP(ARRHO,TT)*PO
00122 30* GBETA = BETA(ARRHO,TT)*T0
00123 31* GCP = CPF(ARRHO,TT)/CP0
00124 32* C
00124 33* A11 = FPR
00125 34* A13 = W*RRHO
00126 35* A31 = GKAPPA
00127 36* A32 = -GBETA
00130 37* A33 = 1.0/W
00131 38* A21 = FZ*(1.0+T*A32)/(RHO*GCP)
00132 39* A23 = W*FMACH/GCP
00132 40* C
00133 41* B1 = (-FR*RRHO*W**2)/2.0
00134 42* B2 = FNU*(TW(NCTL,1)-T)
00135 43* G0 TO (5,10), IFLOW
00135 44* C
00135 45* C1 = A21*A32-A31
00136 46* C2 = A23*A32-A33
00137 47* C3 = A11*C2-A13*C1
00140 48* C
00141 49* DY(1) = (B1*C2-A13*A32+B2)/C3
00142 50* DY(2) = (A11*A32*B2-B1*C1)/C3
00143 51* DY(3) = -(A31*DY(1)+A33*DY(2))/A32
00144 52* RETURN
00144 53* C
00144 54* C
00145 55* C1 = DTHETA(NCTL)*A32
00146 56* C2 = A11*A33-A13*A31
00146 57* C
00147 58* DY(1) = (A13*C1+A33*B1)/C2

```

07  
08

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\*\*\*\*\* DERIVL \*\*\*\*\*  
00150 59\* .DY(2) = -(A11\*C1+A31\*B1)/C2  
00151 60\* RETURN  
00152 61\* END

END OF COMPILATION: . NO DIAGNOSTICS.

SHDG\*P \*\*\*\*\* DERIVM \*\*\*\*\*

QFOR,S ME\*NASAS5,DERIVM,ME\*NASAS5,DERIVM  
 FOR S9A-07/13/72-20:55:28 (0,)

SUBROUTINE DERIVM ENTRY POINT 002331

STORAGE USED: CODE(1) 0023601 DATA(0) 0010611 BLANK COMMON(2) 001115

COMMON BLOCKS:

0003 DVM 000002  
 0004 QIN 001610  
 0005 FLDINL 000457  
 0006 DVC4FL 000002

EXTERNAL REFERENCES (BLOCK, NAME)

0007 CONVEC  
 0010 GRAD  
 0011 ODX  
 0012 D2DX2  
 0013 THCFN  
 0014 DTHCFN  
 0015 CPFN  
 0016 YINT  
 0017 RHOF  
 0020 VISC  
 0021 THCF  
 0022 CPF  
 0023 FLSRT  
 0024 CAPPA  
 0025 BETA  
 0026 THCTB  
 0027 CPTB  
 0030 DTHCTB  
 0031 THCMP  
 0032 CPMP  
 0033 DTHCMP  
 0034 EXP  
 0035 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000041	1146	0001	000047	1216	0001	000061	1256
0001	000220	140L	0001	000111	1426	0001	000124	1466
0001	000204	1706	0001	000243	2056	0001	000263	2106
0001	000341	2316	0001	000366	2406	0001	000401	2476
0001	000575	2776	0001	000772	3366	0001	001121	3666
0001	001174	4056	0001	001211	4186	0001	001235	4206
0001	001377	4466	0001	001414	4546	0001	001434	4626
0001	000633	502L	0001	001534	5066	0001	000760	506L
0001	001731	5276	0001	001755	5376	0001	002116	5356
0001	002177	901L	0001	002215	902L	0001	002221	903L
0024	R	000000	CAPPA	0000	R	000722	CCP	
0001	000075	1336	0001	000173	130L	0001	000173	130L
0001	000146	1606	0001	000141	1546	0001	000141	1546
0001	000323	2266	0001	000310	2176	0001	000310	2176
0001	000536	2716	0001	000420	2536	0001	000420	2536
0001	001154	3776	0001	001137	3716	0001	001137	3716
0001	001361	4436	0001	001272	4316	0001	001272	4316
0001	001475	4756	0001	001451	4716	0001	001451	4716
0001	001056	515L	0001	001055	510L	0001	001055	510L
0001	002112	900L	0001	002262	6056	0001	002262	6056
0025	R	000000	BETA	0002	R	000562	AUX2	
0000	R	000676	CON	0002	R	001107	CON	

```

0000 R 000702 CONJ
0015 R 000000 CPF1
0002 R 001072 DELTA
0014 R 000000 DTHCFN
0002 R 001053 DX1MP
0000 R 000000 DZ1F
0005 R 000456 FOF
0002 R 001046 FOF
0002 R 001111 FRD
0002 R 000036 FT
0000 R 000724 GBETA
0000 R 000723 GKAPA
0000 R 000456 INJ>S
0002 I 001024 LC
0002 I 001011 LL1
0002 I 001016 LL6
0002 I 001007 LTBZ
0002 I 000777 MZ
0002 I 001023 NEQJS
0002 I 001020 NRM>1
0004 I 001604 NRM
0002 I 001071 PI
0004 I 000454 Q1F4
0002 R 000524 QV>3
0000 R 000713 RETR
0002 R 001067 RH00
0002 R 001032 TEV>3
0031 R 000000 THCMP
0004 R 000000 TM
0000 R 000675 TT
0002 R 000050 TW
0000 R 000705 W1F>DT
0002 R 000000 X11J
0000 R 000741 XTHCT
0002 R 001060 XX11
0002 R 001056 XX4
0000 R 000753 X01
0000 R 000751 X4
0002 R 000550 ZETA

0002 R 000740 CONVP
0027 R 000000 CPFB
0002 R 000764 DOB
0030 R 000000 DTHCTB
0000 R 000050 DX1W
0000 R 000524 DZ1TW
0006 R 000001 FLUX1
0002 I 001042 FR
0002 I 001104 FREJ
0002 R 001113 FXHJ
0000 R 000733 GDTHC
0000 I 000671 I
0000 I 000672 K
0002 I 001032 LIM
0002 I 001013 LL3
0002 I 001006 LMP
0002 I 001030 M41
0002 I 001035 NCONV
0002 I 001027 N41
0002 I 001000 NR1B
0000 R 000707 PFLC
0000 R 000714 PRTR
0000 R 000757 QMP
0004 I 000764 Q5TB
0000 R 000710 RHOC
0000 R 000721 RR40
0000 R 000715 TFLIN
0002 I 001077 TI
0002 R 000276 TWP
0000 R 000750 TTHCMP
0002 R 001054 T0
0002 I 001066 W0
0002 R 001054 XRE
0000 R 000741 XVFIS
0000 R 000743 XX2
0000 R 000744 XX5
0000 R 000727 X1
0016 R 000000 YINT

0000 R 000701 CPARM
0002 R 001070 CP0
0000 R 000024 DP
0000 R 000036 DW
0002 R 001051 DZ
0002 I 001103 FENTH
0002 R 001041 FMACH
0002 R 001037 FR
0002 I 001043 FRL
0002 R 001112 FXOH
0000 R 000734 GFNU
0002 I 001003 IFLOW
0000 I 000761 K7
0002 I 001031 LIMWRT
0002 I 001004 LL4
0002 I 001015 LL5
0002 I 001005 LTB
0002 I 001025 NSTOTR
0002 I 001026 NCTM
0002 I 001001 NRMP
0004 I 001605 NSRD
0002 R 001062 PHIM
0000 R 000756 GFN
0002 R 000360 GRFN
0002 I 001074 RDTWRT
0002 R 001114 RHOFN
0000 R 000673 TABS
0013 R 000000 THCFN
0002 R 001076 TINTL
0002 R 001075 TREF
0000 R 000747 TTMBAR
0005 R 000312 WIFLD
0002 R 000005 X1MP
0000 R 000712 XTHCF
0002 R 001057 XX10
0002 R 001055 XX3
0000 R 000716 X00
0000 R 000740 X3
0000 R 000735 Y1

```

SUBROUTINE DERIVM(Y,DY,TIME)  
THIS SUBROUTINE COMPUTES :  
ALL DERIVATIVES OF THE VARIABLE Y WITH RESPECT TO TIME

```

COMMON X1B(5),X1MP(5),FP(10),FW(10),FT(10),TW(10,5),TEMP(10,10),
1 TMP(10,5),GRN(10,10),GRMP(10),X1FN(10),ZETA(10),AUX2(10),
2 CONFN(10,10),CONMP(10),COB(10),D03(10)
COMMON NX,MZ,NR1B,NRMP,NCTL,IFLOW,LT,LTB,LMP,LTBMZ,LTBMZ,LT1,
1 LL2,LL3,LL4,LL5,LL6,LL7,NR1P1,NR1B1,NRMP2,NEQUS,LC,NSTOTR,
2 NCTM,NM1,M41,LIMWRT,L14,M0D,NCCZ,NCONV
COMMON FRE,FR,FNU,FMACH,FPR,FRL,FRN,FFO,FFZ,FRAD,DZ,DX1TB,DX1MP,
1 XRE,XX3,XX4,XX10,XX11,XX12,PHIM,PHIF,T0,P0,W0,RH00,CP0,PI
2 DELTA,RLIMIT,ROTWRT,TREF,TINTL,TI,RTEND,FLUX,GRF,FENTH,

```

```

***** DERIVM *****
00105 15* FREJ, HFN, XL, COH, DXI, FRD, FXOH, FXHM, RHOFN
00106 16* COMMON /DVM/XX20, XX48
00107 17* COMMON /GIN/TM(100), GSFN(100,2), GIPN(100,2), GSTB(100,2),
00108 18* GITB(100,2), NTM, NSRD, QTO, TX
00109 19* COMMON /FLDINL/ NELDIA, LFLO, TMEFLD(100), IFLO(100), WIFLD(100),
00110 20* FLDINT
00111 21* COMMON /DVMFL/ WRAT, FLUXI
00112 22*
00113 23* DIMENSION Y(200), DY(200), DZTF(10), DTW(10), DTW(10), DP(10), DW(10),
00114 24* DXTW(10,10), DZTW(10,10), DZXTW(10,10), DZTTW(10,10),
00115 25*
00116 26* DO 100 J=1, MZ
00117 27* 100 TEMP(J,1) = Y(LL4+J)
00118 28* DO 105 I=2, NX
00119 29* K = MZ*(I-2)
00120 30* DO 105 J=1, MZ
00121 31* 105 TEMP(J,I) = Y(K+J)
00122 32*
00123 33* DO 110 J=1, MZ
00124 34* 110 TMP(J,1) = Y(LL4+J)
00125 35* IF(NRTB.EQ.0) GO TO 130
00126 36* DO 115 I=2, NRMP
00127 37* K = LMP+(I-2)*MZ
00128 38* DO 115 J=1, MZ
00129 39* 115 TMP(J,I) = Y(K+J)
00130 40* DO 120 J=1, MZ
00131 41* TW(J,1) = Y(K)
00132 42* TMP(J, NRMP) = Y(K)
00133 43* DO 120 I=1, NRTBI
00134 44* K = LT8+(I-1)*MZ+J
00135 45* 120 TW(J,I) = Y(K)
00136 46*
00137 47* GO TO 140
00138 48* DO 135 J=1, MZ
00139 49* K = LL4+J
00140 50* TMP(J, NRMP) = Y(K)
00141 51* 135 TEMP(J,1) = Y(K)
00142 52*
00143 53* 140 TABS = TIME*TREF*3600.0
00144 54* TX = TABS/60.0
00145 55* IF(INCONV.GT.0) CALL CONVEC (TABS)
00146 56* CALL GRAD
00147 57*
00148 58* FIN
00149 59*
00150 60*
00151 61* DO 2 I=2, NMI
00152 62* DO 1 J=1, MZ
00153 63* 1 DTW(J) = TEMP(J,1)
00154 64* CALL DDZ (DTW, DP, DZ, MZ)
00155 65* CALL DZDZ2(DTWF, DW, DZ, MZ)
00156 66* DO 2 J=2, NMI
00157 67* 2 DZTW(J,1) = DP(J)
00158 68* DTW(J,1) = DW(J)
00159 69* DZTTW(J,1) = DW(J)
00160 70*
00161 71* DO 4 J=2, NMI

```

\*\*\*\*\* DERIVM \*\*\*\*\*

```

00230 DO 3 I=1,NX
00235 DTF(I) = TEMP(J,I)
00236 CALL DDX (DTF,DP,DXI,NX)
00237 CALL D2DX2(DTF,DW,DXI,NX)
00242 DO 4 I=2,NMI
00243 DXTW(J,I) = DP(I)
00244 D2XTW(J,I) = DW(I)
00245 C
00246 DO 9 I=2,NMI
00251 IM2 = I-2
00252 DO 9 J=2,MM1
00255 K = IM2*MZ+J
00256 TT = T0+TEMP(J,I)
00257 CON = THCFN(TT)
00260 DCON = DTHCFN(TT)*T0
00261 DIFF = CON/(RHOFN*CPFN(TT))*FXHW
00262 CPARM = FRD/CON
00262 88* C
00262 89* C
00263 COND = DXTW(J,I)+FXOH*D2ZTW(J,I)-COB(I)*DXTW(J,I)
00263 91* +DCON*(DXTW(J,I)**2+FXOH*D2ZTW(J,I)**2)
00264 92* 9 DY(K) = DIFF*(COND-DOB(I))*CPARM*(GRFN(J,I)-CONFN(J,I))
00264 93* C
00264 94* C
00264 95* C
00264 96* TIP OF FIN
00267 NM2 = NX-2
00270 DO 11 J=2,MM1
00273 K = NM2*MZ+J
00274 11 DY(K) = (18.*DY(K-MZ)-9.*DY(K-2*MZ)+2.*DY(K-3*MZ))/11.
00274 100* C
00274 101* C
00274 102* C
00274 103* C
00274 104* C
00274 105* C
00274 106* C
00274 107* C
00276 500 FT(I) = Y(I+LT)
00301 C
00301 109* C
00303 IF (LFLD.EQ.1) GO TO 502
00305 XTME = TIME*TREF
00306 FT(1) = YINT(TMEFLD,TIFLD,NFLDTA,3,XTME)/T0
00307 FLDINT = FT(1)
00310 WIFLOT = YINT(TMEFLD,WIFLD,NFLDTA,3,XTME)
00311 WRAT = WIFLOT/FLUXI
00311 115* C
00311 116* C
00312 502 TFLC = FT(LC)*T0
00313 PFLC = FP(LC)*P0
00314 RHOC = RHOF(PFLC,TFLC)
00315 XVISC = VISC(RHOC,TFLC)
00316 XTHCF = THCF(RHOC,TFLC)
00317 RETR = XRE/XVISC
00320 PRTR = (XVISC*CPF(RHOC,TFLC))/XTHCF
00321 TFLIN = Y(LI+1)
00322 CALL FLSTRT(RETR,PRTR,DELTA)
00323 X00 = DELTA*RETR*PRTR*XTHCF/4.0
00324 CALL DDX(FW,DW,DZ,MZ)
00324 126* C
00324 127* C
00324 128* C

```

\*\*\*\*\* DERIV \*\*\*\*\*

```

00325 129*
00325 130*
00326 131*
00327 132*
00331 133*
00333 134*
00334 135*
00334 136*
00335 137*
00340 138*
00341 139*
00342 140*
00343 141*
00344 142*
00344 143*
00345 144*
00346 145*
00347 146*
00350 147*
00351 148*
00352 149*
00354 150*
00355 151*
00356 152*
00357 153*
00360 154*
00360 155*
00361 156*
00361 157*
00361 158*
00361 159*
00361 160*
00363 161*
00365 162*
00370 163*
00373 164*
00375 165*
00376 166*
00401 167*
00403 168*
00404 169*
00407 170*
00410 171*
00410 172*
00413 173*
00416 174*
00417 175*
00422 176*
00423 177*
00424 178*
00425 179*
00430 180*
00433 181*
00434 182*
00435 183*
00436 184*
00437 185*

C CALL DD(XFT,DTF,DZ,MZ)
  DY(LT+1) = 0.0
  IF (LFLD.EQ.2) GO TO 506
  IF (TIME.LE.2.5) DY(LT+1) = 7.0*EXP(-7.0*TIME)*(1.0-TINTL)
  506 AUX2(1) = FNU*(TW(1,1))-FT(1)
  AUX2(1) = -X00*AUX2(1)

C DO 520 I=2,MZ
  PP = P0*FP(I)
  TT = T0*FT(I)
  RHO = 1.0/FW(I)
  RRHO = RHO0*RHO
  CCP = CPF(RRHO,TT)

C GKAPPA = CAPP( RRHO,TT)*P0
  GBETA = BETA( RRHO,TT)*T0
  GCP = CCP/CP0
  AUX2(1) = FNU*(TW(1,1))-FT(1)
  LLT = LT+1
  IF (GKAPPA.LT. 1.0E-08) GO TO 510
  X1 = FZ*GBETA*FT(1)/(GKAPPA*GCP*RHO)
  GO TO 515
  510 X1 = 0.0
  515 X2 = 1.0-X1*GBETA
  DY(LLT) = (AUX2(1)+FMACH*FR*(FW(I)**2)/2.0)/(RHO*GCP)-X1*DW(I))/
  1 X2-FW(I)*DTF(I)
  520 AUX2(1) = -X00*AUX2(1)

C TUBE WALL TEMPERATURE
  IF (NRTB.EQ.0) GO TO 900
  DO 625 J=1,MZ
  DO 615 I=1,NRTB
  DTF(I) = TW(J,I)
  615 CALL DD(XDTF,D2TF,DXITB,NRTB)
  DO 620 I=1,NRTB
  DP(I) = XITB(I)*D2TF(I)
  CALL DD(OP,DTF,DXITB,NRTB)
  DO 625 I=2,NRTB1
  D2TW(J,I) = DTF(I)/XITB(I)
  625 DXTW(J,I) = (D2TF(I))**2

C DO 630 I=2,NRTB1
  K = LIB*(I-1)*MZ
  DO 630 J=1,MZ
  TTB = TW(J,I)*T0
  GFO = THCTB(TTB)/CRTB(TTB)*FFO
  GOTHC = T0*DTHCTB(TTB)
  630 DY(K+J) = GFO*(D2XTW(J,I)+GOTHC*(DXTW(J,I)))
  DO 635 J=1,MZ
  ITB = TW(J,1)*T0
  GFNU = XTHCF*FNU/(4.0*THCTB(TTB))*RETR*PRTR*DELTA*DXITB
  Y1 = TW(J,1)-FT(J)
  X1 = 4.0*DY(LTBZ+J)-DY(LTB2MZ+J)
  X2 = 3.0*GFNU*(1.0-T0*DTHCTB(TTB))*Y1

```



\*\*\*\*\* DERIVM \*\*\*\*\*

00440 186\* 635 DY(LTB+J) = (X1+GFNU\*DY(LT+J))/X2

00440 187\* C  
 00440 188\* C  
 00440 189\* C  
 00440 190\* C  
 00440 191\* C  
 00440 192\* C  
 00440 193\* C

PROTECTION LAYER TEMPERATURE

00442 194\* DO 730 J=1,MZ  
 00445 195\* DO 720 I=1,NRMP  
 00450 196\* DO 720 DTF(I) = TMP(J,I)  
 00452 197\* CALL DDX(DTF,D2TF,DXIMP,NRMP)  
 00453 198\* DO 725 I=1,NRMP  
 00456 199\* DO 725 DP(I) = XIMP(I)\*D2TF(I)  
 00460 200\* CALL DDX(DP,D2DP,DXIMP,NRMP)  
 00461 201\* DO 730 I=2,NRMP1  
 00464 202\* D2XTW(J,I) = DTF(I)/XIMP(I)  
 00465 203\* DO 730 DXTW(J,I) = D2TF(I)\*\*2

00465 203\* C  
 00470 204\* DO 735 I=2,NRMP1  
 00473 205\* K = LMP+(I-2)\*MZ  
 00474 206\* DO 735 J=1,MZ  
 00477 207\* TTM = TMP(J,I)\*T0  
 00500 208\* GFO = THCMP(TTM)/CPMP(TTM)\*FOF  
 00501 209\* GDTHC = T0\*DTHCMP(TTM)  
 00502 210\* 735 DY(K+J) = GFO\*(D2XTW(J,I)+6DTHC\*(DXTW(J,I)))  
 00502 211\* C  
 00502 212\* C  
 00502 213\* C  
 00502 214\* C  
 00502 215\* C  
 00505 216\* C  
 00510 217\* C  
 00511 218\* C  
 00512 219\* C  
 00513 220\* C  
 00514 221\* C  
 00515 222\* C  
 00516 223\* C  
 00517 224\* C  
 00520 225\* C  
 00520 226\* C

FIN - TUBE - PROTECTION LAYER INTERFACE

00521 227\* YY1 = 4.0\*DY(LL1+J)-DY(LL2+J)  
 00522 228\* YY1 = YY1+XX1\*XX3\*(4.0\*DY(LMP+J)-DY(LL3+J))  
 00523 229\* YY1 = YY1+XX2\*XX4\*(4.0\*DY(J)-DY(MZ+J))  
 00524 230\* 740 DY(LL4+J) = YY1/(3.0+XX1\*(3.0\*XX3-XX5)+XX2\*(3.0\*XX4-XX6))  
 00524 231\* C  
 00524 232\* C  
 00524 233\* C  
 00524 234\* C  
 00524 235\* C  
 00526 236\* C  
 00531 237\* C  
 00532 238\* C  
 00533 239\* C  
 00534 240\* C  
 00534 241\* C  
 00534 242\* C

MANIFOLD-FIN INTERFACE

00526 236\* DO 750 I=1,NW1  
 00531 237\* K = I\*MZ-(MZ-1)  
 00532 238\* DY(K) = DY(LL4+1)  
 00533 239\* K = I\*MZ  
 00534 240\* 750 DY(K) = DY(LL4+MZ)  
 00534 241\* C  
 00534 242\* C

OUTER BOUNDARY OF PROTECTION LAYER

```

***** DERIV *****
00534 243* C
00534 244* C
00534 245* C
00534 246* C
00534 247* C
00536 248*
00541 249*
00542 250*
00543 251*
00544 252*
00545 253*
00546 254*
00546 255*
00547 256*
00550 257*
00551 258*
00551 259*
00553 260*
00553 261*
00554 262*
00557 263*
00560 264*
00561 265*
00562 266*
00563 267*
00564 268*
00565 269*
00567 270*
00570 271*
00571 272*
00573 273*
00573 274*
00574 275*
00575 276*
00576 277*
00577 278*
00600 279*
00600 280*
00602 281*
00603 282*
00604 283*
00607 284*
00610 285*
00611 286*
00612 287*
00614 288*
00615 289*

DO 810 J=1,MZ
TTMBAR = (8.0*TMP(J, NRMP)+5.0*TMP(J, NRMP1)-TMP(J, NRMP2))*T0/12.
TTHCMP = TTHCMP(TTM)
GFO = TTHCMP*FOF/CPMP(TTMBAR)
GOTHC = T0*OTHCMP(TTM)
X1 = TMP(J, NRMP)-TMP(J, NRMP1)
X2 = 1.0-GOTHC*(3.0*TMP(J, NRMP)-4.0*TMP(J, NRMP1)+
1 X4 = TMP(J, NRMP2))/4.0
X4 = FRAD*XX12*(GRMP(J)+CONMP(J))/TTHCMP
X5 = -X1*XX11*X2-X4
810 DY(LL5+J) = (12.0*X5*GFO-5.0*DY(LL6+J)+DY(LL7+J))/8.0
RETURN
C
C
900 DO 905 J=1,MZ
K = LL4+J
TT = Y(K)*T0
TFLC = T0*FT(J)
RHOC = RH00/FW(J)
X01 = XX10*CPFB(TT)+XX11*CPMP(TT)+XX12*CPFN(TT)
X03 = XX4*THCFN(TT)
IF(J.NE.1) GO TO 901
TMPGRD = (5.0*(TEMP(2,2)-TEMP(2,1))+TEMP(3,1)-TEMP(3,2))/XX48
GO TO 903
901 IF(J.NE.MZ) GO TO 902
TMPGRD = (5.0*(TEMP(MM1,2)-TEMP(MM1,1))+TEMP(LL6,1)
1 -TEMP(LL6,2))/XX48
GO TO 903
902 TMPGRD = (TEMP(J,2)-TEMP(J,1))/DXI
903 GFN = GRFN(J,1)+CONFN(J,1)
GMP = GRMP(J)+CONMP(J)
905 DY(K) = (AUX2(J)-FRAD*(GMP+XX20*GFN)+X03*TMPGRD)/X01
C
K1 = LL4+1
K2 = LL4+MZ
DO 910 I=1,NM1
K = I*MZ
DY(K) = DY(KZ)
K = K-MZ+1
910 DY(K) = DY(K1)
RETURN
END

```

END OF COMPILATION: NO DIAGNOSTICS.

GH06/P \*\*\*\*\* DTHCFN/FNAL \*\*\*\*\*

\*\*\*\*\* DTHCFN/FNAL \*\*\*\*\*  
QFOR,S ME\*NASAS5.DTHCFN/FNAL,ME\*NASAS5.DTHCFN/FNAL  
FOR S9A-07/13/72-20:55:36 (0,)

FUNCTION DTHCFN ENTRY POINT 000033

STORAGE USED: CODE(1) 000040; DATA(0) 000017; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 THCFN  
0004 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000006 DT 0000 R 000011 DTHC 0000 R 000000 DTHCFN 0000 R 000010 THC  
0003 R 000000 T-HCFN 0000 R 000007 THETA 0000 R 000005 T0 0000 R 000001 X1  
0000 R 000003 X3 0000 R 000004 X4 0000 R 000002 X2

FUNCTION DTHCFN (T)

00101 1\* C  
00101 2\* C  
00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
00101 4\* C (1/K)(DK/DI) AS A FUNCTION OF TEMPERATURE (R) OF ALUMINUM 7075  
00101 5\* C UNITS (1/R)  
00101 6\* C TEMP. 6E. 300.0.AND.LE. 1200.0 R  
00101 7\* C  
00101 8\* C DATA X1,X2,X3,X4 /13.0665/0.6655\*-5.25\*1.015/T0\*DT /400.0\*200.0/  
00101 9\* C THETA = (T-T0)/DT  
00112 10\* C THC = THCFN(T)  
00113 11\* C DTHC = ((X4\*THETA+X3)\*THETA+X2)\*THETA+X1)/DT  
00114 12\* C DTHCFN = DTHC/THC  
00115 13\* C RETURN  
00116 14\* C  
00117 END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* DTHCFN/FNBR \*\*\*\*\*

\*\*\*\*\* DTHCFN/FNBR \*\*\*\*\*

GFORP5 ME\*NASAS.DTHCFN/FNBR,ME\*NASAS.DTHCFN/FNBR  
FOR S9A-07/13/72-20:55:39 (0,)

FUNCTION DTHCFN ENTRY POINT 000033

STORAGE USED: CODE(1) 000040; DATA(0) 000017; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 THCMP  
0004 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000006 DT 0000 R 000011 DTHC 0000 R 000000 DTHCFN 0000 000012 INJPS 0000 R 000010 THC  
0003 R 000000 THCMP 0000 R 000007 THETA 0000 R 000005 T0 0000 R 000001 X1 0000 R 000002 X2  
0000 R 000003 X3 0000 R 000004 X4

FUNCTION DTHCFN(T)

00101 1\*  
00101 2\*  
00101 3\*  
00101 4\*  
00101 5\*  
00101 6\*  
00101 7\*  
00101 8\*  
00103 9\*  
00103 10\*  
00112 11\*  
00113 12\*  
00114 13\*  
00115 14\*  
00116 15\*  
00117 16\*

THIS FUNCTION SUBPROGRAM COMPUTES:  
(1/K)(DK/DT) AS A FUNCTION OF TEMPERATURE (R) OF BERYLLIUM WITH  
0.84-1.68 BEQ

UNITS (1/R)  
TEMP. GE. 400.0 AND LE. 1700.0 R

DATA X1,X2,X3,X4 /-10.5643,1.65366,-0.60501,0.080668/T0,DT /400.0  
1,200.0/

THETA = (T-T0)/DT

THC = THCMP(T)

DTHC = ((X4\*THETA+X3)\*THETA+X2)\*THETA+X1)/DT

DTHCFN = DTHC/THC

RETURN

END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG:P \*\*\*\*\* DTHCFN/FNCU \*\*\*\*\*

\*\*\*\*\* DTHCFN/FNCJ \*\*\*\*\*

FOR S ME\*NASAS.DTHCFN/FNCU,ME\*NASAS.DTHCFN/FNCU  
FOR S9A-07/13/72-20:55:41 (07)

FUNCTION DTHCFN ENTRY POINT 000030

STORAGE USED: CODE(1) 000035; DATA(0) 000015; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 THCTB  
0004 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000004 DT 0000 R 000007 DTHC 0000 R 000000 DTHCFN 0000 R 000006 THC  
0003 R 000000 THCTB 0000 R 000005 THETA 0000 R 000003 T0 0000 R 000001 X1 0000 R 000002 X2

00101 1\* FUNCTION DTHCFN(T)  
 00101 2\* C  
 00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
 00101 4\* C (1/K)(DK/DT) AS A FUNCTION OF TEMPERATURE (R) OF COPPER  
 00101 5\* C UNITS (1/R)  
 00101 6\* C TEMP. GE. 500.0 AND LE. 1800.0 R  
 00101 7\* C  
 00103 8\* DATA X1\*X2 /-2.62067\*-0.121/\*T0\*DT /600.0\*200.0/  
 00110 9\* THETA = (T-T0)/DT  
 00111 10\* THC = THCTB(T)  
 00112 11\* DTHC = (X2\*THETA\*THETA\*X1)/DT  
 00113 12\* DTHCFN = DTHC/THC  
 00114 13\* RETURN  
 00115 14\* END

END OF COMPILATION: NO DIAGNOSTICS.

END6.P \*\*\*\*\* DTHCMP/MPAL \*\*\*\*\*

DFOR,S ME\*NASAS,DTHCMP/MPAL,ME\*NASAS,DTHCMP/MPAL  
FOR S9A-07/13/72-20:55:44 (0\*)

FUNCTION DTHCMP ENTRY POINT 000033

STORAGE USED: CODE(1) 000040; DATA(0) 000017; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 THCFN  
0004 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000006 DT 0000 R 000011 DTHC 0000 R 000000 DTHCMP 0000 000012 INJPS 0000 R 000010 THC  
0003 R 000000 THCFN 0000 R 000007 THETA 0000 R 000005 T0 0000 R 000001 X1 0000 R 000002 X2  
0000 R 000003 X3 0000 R 000004 X4

00101 1\* FUNCTION DTHCMP(T)  
00101 2\* C  
00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
00101 4\* C (1/K)(DK/DT) AS A FUNCTION OF TEMPERATURE (R) OF ALUMINUM 7075  
00101 5\* C UNITS (1/R)  
00101 6\* C TEMP. GE. 300.0,AND,LE. 1200.0 R  
00101 7\* C  
00101 8\* DATA X1,X2,X3,X4 /13.0665,0.6655,-5.25,1.015/.T0\*DT /400.0,200.0/  
00112 9\* THETA = (T-T0)/DT  
00113 10\* THC = THCFN(T)  
00114 11\* DTHC = ((X4\*THETA+X3)\*THETA+X2)\*THETA+X1)/DT  
00115 12\* DTHCMP = DTHC/THC  
00116 13\* RETURN  
00117 14\* END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* DTHCMP/MPBR \*\*\*\*\*

\*\*\*\*\* DTHCMP/MPBR \*\*\*\*\*

QFOR,S ME\*NASAS.DTHCMP/MPBR,ME\*NASAS.DTHCMP/MPBR  
FOR S9A-07/13/72-20:55:46 (0,)

FUNCTION DTHCMP ENTRY POINT 000033

STORAGE USED: COJE(1) 000040; DATA(0) 000017; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 THCMP  
0004 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000006 D1 0000 R 000011 DTHC 0000 R 000000 DTHCMP 0000 R 000010 THC  
0003 R 000000 THCMP 0000 R 000007 THETA 0000 R 000005 T0 0000 R 000001 X1 0000 R 000002 X2  
0000 R 000003 X3 0000 R 000004 X4

00101 1\* FUNCTION DTHCMP (T)

00101 2\* C

00101 3\* C

00101 4\* C

00101 5\* C

00101 6\* C

00101 7\* C

00101 8\* C

00103 9\*

00103 10\*

00112 11\*

00113 12\*

00114 13\*

00115 14\*

00116 15\*

00117 16\*

THIS FUNCTION SUBPROGRAM COMPUTES:  
(1/K)(DK/DT) AS A FUNCTION OF TEMPERATURE (R) OF BERYLLIUM WITH  
0.84-1.68 BEQ

UNITS (1/R)

TEMP. GE. 400.0 AND LE. 1700.0 R

DATA X1,X2,X3,X4 /-10.5643,1.65366,-0.60501,0.080658/,T0,DT /400.0  
1,200.0/

THETA = (T-T0)/DT

THC = THCMP(T)

DTHC = ((X4\*THETA+X3)\*THETA+X2)\*THETA+X1)/DT

DTHCMP = DTHC/THC

RETURN

END

END OF COMPILATION: NO DIAGNOSTICS.

CHDG,P \*\*\*\*\* DTHCMP/MPCU \*\*\*\*\*

FOR S ME\*NASAS.DTHCMP/MPCU,ME\*NASAS.DTHCMP/MPCU  
 FOR S9A-07/13/72-20:55:50 (0.)

FUNCTION DTHCMP ENTRY POINT 000030

STORAGE USED: CODE(1) 000035; DATA(0) 000015; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 THCTB  
 0004 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000004 DT 0000 R 000007 DTHC 0000 R 000000 DTHCMP 0000 R 000006 THC  
 0003 R 000000 THCTB 0000 R 000005 THETA 0000 R 000003 T0 0000 R 000001 X1  
 0000 R 000002 X2

FUNCTION DTHCMP(T)

00101 1\* C  
 00101 2\* C  
 00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
 00101 4\* C (1/K)(DK/DT) AS A FUNCTION OF TEMPERATURE (R) OF COPPER  
 00101 5\* C UNITS (1/R)  
 00101 6\* C TEMP. GE. 500.0 AND LE. 1800.0 R  
 00101 7\* C  
 00103 8\* C  
 00110 9\* C  
 00111 10\* C  
 00112 11\* C  
 00113 12\* C  
 00114 13\* C  
 00115 14\* C

DATA X1,X2 /-2.62067,-0.121/,T0,DT /600.0,200.0/  
 THETA = (T-T0)/DT  
 THC = THCTB(T)  
 DTHC = (X2\*THETA\*THETA+X1)/DT  
 DTHCMP = DTHC/THC  
 RETURN  
 END

END OF COMPILATION: NO DIAGNOSTICS.

ENDG:P \*\*\*\*\* DTHCTB/TBAL \*\*\*\*\*



FOR>S ME\*NASAS.DTHCTB/TBAL,ME\*NASAS.DTHCTB/TBAL  
FOR S9A-07/13/72-20:55:53 (0,)

FUNCTION DTHCTB ENTRY POINT 000033

STORAGE USED: CODE(1) 000040; DATA(0) 000017; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 THCFN  
0004 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000006 DT 0000 R 000011 DTHC 0000 R 000000 DTHCTB 0000 000012 INJPS 0000 R 000010 THC  
0003 R 000000 THCFN 0000 R 000007 THETA 0000 R 000005 T0 0000 R 000001 X1 0000 R 000002 X2  
0000 R 000003 X3 0000 R 000004 X4 TEMP. GE. 300.0.AND.LE. 1200.0 R

00101 1\* FUNCTION DTHCTB(T)

00101 2\* C

00101 3\* C

00101 4\* C

00101 5\* C

00101 6\* C

00101 7\* C

00103 8\* C

00112 9\* C

00113 10\* C

00114 11\* C

00115 12\* C

00116 13\* C

00117 14\* C

THIS FUNCTION SUBPROGRAM COMPUTES :  
(1/K)(DK/DT) AS A FUNCTION OF TEMPERATURE (R) OF ALUMINUM 7075  
UNITS (1/R)  
TEMP. GE. 300.0.AND.LE. 1200.0 R  
DATA X1,X2,X3,X4 /13.0665,0.6655,-5.25,1.015/,T0,DT /400.0,200.0/  
THETA = (T-T0)/DT  
THC = THCFN(T)  
DTHC = (((X4\*THETA+X3)\*THETA+X2)\*THETA+X1)/DT  
DTHCTB = DTHC/THC  
RETURN  
END

END OF COMPILATION: NO DIAGNOSTICS.

END OF P \*\*\*\*\* DTHCTB/TBRR \*\*\*\*\*

FOR S ME\*NASAS.DTHCTB/TBBR,ME\*NASAS.DTHCTB/TBBR  
 FOR S9A-07/13/72-20:55:56 (0,)

FUNCTION DTHCTB ENTRY POINT 000033

STORAGE USED: CODE(1) 000040; DATA(0) 000017; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 THCMP  
 0004 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000006 DT 0000 R 000011 DTHC 0000 R 000000 DTHCTB 0000 000012 INJPS 0000 R 000010 THC  
 0003 R 000000 THCMP 0000 R 000007 THETA 0000 R 000005 T0 0000 R 000001 X1 0000 R 000002 X2  
 0000 R 000003 X3 0000 R 000004 X4

00101 1\* FUNCTION DTHCTB(T)  
 00101 2\* C  
 00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
 00101 4\* C (1/K)(DK/DT) AS A FUNCTION OF TEMPERATURE (R) OF BERYLIUM WITH  
 00101 5\* C 0.84-1.68 BEO  
 00101 6\* C UNITS (1/R)  
 00101 7\* C TEMP, GE. 400.0 AND LE. 1700.0 R  
 00101 8\* C  
 00103 9\* DATA X1,X2,X3,X4 /-10.5643,1.65366,-0.60501,0.080668/,T0,DT /400.0  
 00103 10\* 1,200.0/  
 00112 11\* THETA = (T-T0)/DT  
 00113 12\* THC = THCMP(T)  
 00114 13\* DTHC = ((X4\*THETA+X3)\*THETA+X2)\*THETA+X1)/DT  
 00115 14\* DTHCTB = DTHC/THC  
 00116 15\* RETURN  
 00117 16\* END

END OF COMPILATION: NO DIAGNOSTICS.

ENDG,P \*\*\*\*\* DTHCTB/TBCU \*\*\*\*\*

\*\*\*\*\* DTHCTB/TBCU \*\*\*\*\*

QFOR: S ME\*NASAS.DTHCTB/TBCU, ME\*NASAS.DTHCTB/TBCU  
FOR S9A-07/13/72-20:55:58 (0,)

FUNCTION DTHCTB ENTRY POINT 000030

STORAGE USED: CODE(1) 000035; DATA(0) 000015; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 THCTB  
0004 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000004 DT 0000 R 000007 DTHC 0000 R 000000 DTHCTB 0000 R 000006 THC  
0003 R 000000 THCTB 0000 R 000005 THETA 0000 R 000003 T0 0000 R 000001 X1  
0000 R 000002 X2

00101 1\* FUNCTION DTHCTB (T)

00101 2\* C  
00101 3\* C  
00101 4\* C  
00101 5\* C  
00101 6\* C  
00101 7\* C  
00101 8\* C  
00103 9\*  
00110 10\*  
00111 11\*  
00112 12\*  
00113 13\*  
00114 14\*  
00115

THIS FUNCTION SUBPROGRAM COMPUTES :  
(1/K)(DK/DT). AS A FUNCTION OF TEMPERATURE (R) OF COPPER  
UNITS (1/R)

TEMP. GE. 500.0 AND LE. 1800.0 R

DATA X1,X2 /-2.62067,-0.121/,T0,DT /600.0,200.0/

THETA = (T-T0)/DT

THC = THCTB(T)

DTHC = (X2\*THETA\*THETA+X1)/DT

DTHCTB = DTHC/THC

RETURN  
END

END OF COMPILATION: NO DIAGNOSTICS.

DHD6.P \*\*\*\*\* D2DX2 \*\*\*\*\*

\*\*\*\*\* D2DX2 \*\*\*\*\*

QFOR,S ME\*NASAS.D2DX2,ME\*NASAS.D2DX2  
FOR 59A-07/13/72-20:56:00 (0,)

SUBROUTINE D2DX2 ENTRY POINT 000137

STORAGE USED: CODE(1) 000163; DATA(0) 000046; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDJS  
0004 NIO2\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000064 1126 0001 000112 2L 0000 000003 3F 0000 R 000000 DX2 0000 I 000002 I  
0000 000016 INJPS 0000 I 000001 N1

```

00101 1* SUBROUTINE D2DX2 (Y,D2Y,DX,N) 1
00101 2* C
00101 3* C THIS SUBROUTINE COMPUTES :
00101 4* C THE SECOND DERIVATIVE OF THE EQUALLY-SPACED ARRAY Y(N)
00101 5* C
00103 6* DIMENSION Y(N),D2Y(N) 2
00104 7* IF (N.LT.4) GO TO 2 3
00106 8* DX2 = DX**2 4
00107 9* D2Y(1) = (2.0*Y(1)-5.0*Y(2)+4.0*Y(3)-Y(4))/DX2 5
00110 10* N1 = N-1 6
00111 11* DO 1 I=2,N1 7
00114 12* 1 D2Y(I) = (Y(I+1)-2.0*Y(I)+Y(I-1))/DX2 8
00116 13* D2Y(N) = (2.0*Y(N)-5.0*Y(N-1)+4.0*Y(N-2)-Y(N-3))/DX2 9
00117 14* RETURN 10
00120 15* 2 WRITE (6,3) 11
00122 16* 3 FORMAT (27HD2DX2 NOT ENOUGH NODAL PTS.,/) 12
00123 17* RETURN 13
00124 18* END 14

```

END OF COMPILATION: NO DIAGNOSTICS.

QHD6,P \*\*\*\*\* EFFICY \*\*\*\*\*

FOR S ME\*NASAS.EFFICY,ME\*NASAS.EFFICY  
FOR S9A-07/13/72-20:56:02 (0,)

SUBROUTINE EFFICY ENTRY POINT 000134

STORAGE USED: CODE(1) 000161; DATA(0) 000131; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 POLY  
0004 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000007	1166	0001	000011	1216	0001	000040	1326	0001	000042	1356	0001	000073	1466	
0000	R	000064	A	0000	R	000104	AK	0000	R	000073	APR	0000	R	000050	COEF
0000	R	000055	COEFF	0000	R	000043	COF	0000	I	000102	I	0000	I	000103	J
0000	I	000105	K	0003	R	000000	POLY								

```

00101 1* SUBROUTINE EFFICY (NC,NCP,TSOTB,IB,ETA,ETAPR)
00102 2* DIMENSION C(5,7),COF(5),COEF(5),COEFF(7),A(7),APR(7)
00103 3* REAL NC,NCP
00104 4* DATA(C(1,2),I=1,5)/-1.1632,.1633405,-1.629484,3.735489,-2.75355/,
00105 5* 1 (C(1,3),I=1,5)/1.438967,-.5483718,5.091788,-11.26815,7.905002/,
00106 6* 2 (C(1,4),I=1,5)/-1.158833,.7783245,-6.866955,14.98945,-10.35581/,
00107 7* 3 (C(1,5),I=1,5)/.537,-.5282332,4.460133,-9.675815,6.65502/,
00108 8* 4 (C(1,6),I=1,5)/-.1294667,.1666654,-1.360822,2.94164,-2.020815/,
00109 9* 5 (C(1,7),I=1,5)/.01253334,-.01962952,.1562953,-.3370346,.2314797/
00110 10* A(1) = 1.0
00111 11* DO 20 I=2,7
00112 12* DO 10 J=1,5
00113 13* 10 COEF(J) = C(J,I)
00114 14* 20 A(I) = POLY (5,COEF,TSOTB)
00115 15* ETA = POLY (7,A,NC)
00116 16* APR(1) = 0.0
00117 17* DO 40 I=2,7
00118 18* DO 30 J=1,5
00119 19* AK = J-1
00120 20* COF(J) = AK*C(J,I)
00121 40 APR(I) = -(1.0/IB)*POLY(5,COF,TSOTB)
00122 22* COEFF(7) = APR(7)*NC**4
00123 23* DO 50 I=1,6
00124 24* K = I+1
00125 25* AK = I
00126 50 COEFF(I) = APR(I)*AK*(K)*NCP
00127 27* ETAPR = POLY (7,COEFF,NC)
00128 28* RETURN
00129 29* END

```

END OF COMPILATION: NO DIAGNOSTICS.

\*\*\*\*\* EFFICY \*\*\*\*\*  
OHOG/P \*\*\*\*\* ELAS/MPAL \*\*\*\*\*

QFOR,S ME\*NASAS.ELAS/MPAL,ME\*NASAS.ELAS/MPAL  
 FOR S9A-07/13/72-20:56:06 (0,)

FUNCTION ELAS ENTRY POINT 000055

STORAGE USED: CODE(1) 000061; DATA(0) 000040; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDUS  
 0004 NIO25  
 0005 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000010 IF 0000 R 000006 DT 0000 R 000000 ELAS 0000 000032 INJPS 0000 R 000007 THETA  
 0000 R 000005 TO 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3 0000 R 000004 X4

```

00101 1* C FUNCTION ELAS(T)
00101 2* C
00101 3* C THIS FUNCTION SUBPROGRAM COMPUTES :
00101 4* C MODULUS OF ELASTICITY AS A FUNCTION OF TEMPERATURE (R) OF
00101 5* C ALUMINUM 7075
00101 6* C UNITS LBF/SQ.FT
00101 7* C TEMP. GE. 300.0.AND.LE. 1200.0 R
00101 8* C
00103 9* C DATA X1,X2,X3,X4 /10.71,-0.63,-0.115,-0.06/,T0*DT /459.67,200.0/
00112 10* C THETA = (T-T0)/DT
00113 11* C ELAS = ((X4*THETA+X3)*THETA+X2)*THETA+X1)*1.0E06
00114 12* C ELAS = ELAS*144.0
00115 13* C IF (T.GE.300.0.AND.T.LE.1200.0) RETURN
00117 14* C WRITE (6,1) T
00122 15* C 1 FORMAT (1H0,59HMODULUS OF ELASTICITY OF ALUMINUM, TEMP. OUT OF RAN
00122 16* C 1GE, T = ,F8.3,/)
00123 17* C RETURN
00124 18* C END
    
```

END OF COMPILATION: NO DIAGNOSTICS.

QHD6,P \*\*\*\*\* ELAS/MPBR \*\*\*\*\*

FOR'S ME\*NASAS.ELAS/MPBR,ME\*NASAS.ELAS/MPBR  
FOR S9A-07/13/72-20:56:10 (0,)

FUNCTION ELAS ENTRY POINT 000055

STORAGE USED: CODE(1) 000061; DATA(0) 000040; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDUS\$  
0004 NIO2\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000010 IF 0000 R 000006 DT 0000 R 000000 ELAS 0000 000032 INJPS 0000 R 000007 THETA  
0000 R 000005 T0 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3 0000 R 000004 X4

00101 1\* FUNCTION ELAS(T)

00101 2\* C  
00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
00101 4\* C MODULUS OF ELASTICITY AS A FUNCTION OF TEMPERATURE (R) OF  
00101 5\* C BERYLIUM WITH 0.84--1.68 BEQ  
00101 6\* C UNITS LBF/SG.FT  
00101 7\* C TEMP. GE. 400.0 AND LE. 1700.0 R  
00101 8\* C  
00103 9\* DATA X1,X2,X3,X4 /4.36,-3.755,0.335,-0.53/\*T0,DT /459.67,400.0/  
00112 10\* THETA = (T-T0)/DT  
00113 11\* ELAS = (((X4\*THETA+X3)\*THETA+X2)\*THETA+X1)\*1.0E06  
00114 12\* ELAS = ELAS\*144.0  
00115 13\* IF (T.GE.400.0.AND.T.LE.1700.0) RETURN  
00117 14\* WRITE (6,1) T  
00122 15\* 1 FORMAT (1H0,59HMODULUS OF ELASTICITY OF BERYLIUM, TEMP. OUT OF RAN  
00122 16\* 1GE, T = ,F8.3//)  
00123 17\* RETURN  
00124 18\* END

END OF COMPILATION: NO DIAGNOSTICS.

ENDG.P \*\*\*\*\* ELAS/MPCU \*\*\*\*\*



\*\*\*\*\* ELAS/MPCU \*\*\*\*\*

QFOR,S ME\*NASAS.ELAS/MPCU\*ME\*NASAS5.ELAS/MPCU  
FOR 59A-07/13/72-20:56:13 (0\*)

FUNCTION ELAS ENTRY POINT 000055

STORAGE USED: CODE(1) 000061; DATA(0) 000040; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDUS  
0004 NIO2\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000010 IF 0000 R 000006 DT 0000 R 000000 ELAS 0000 000032 INJPS 0000 R 000007 THETA  
0000 R 000005 TO 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3 0000 R 000004 X4

00101 1\* FUNCTION ELAS(T)

```

00101 2* C
00101 3* C THIS FUNCTION SUBPROGRAM COMPUTES :
00101 4* C MODULUS OF ELASTICITY AS A FUNCTION OF TEMPERATURE (R) OF
00101 5* C COPPER
00101 6* C UNITS LBF/50.FT
00101 7* C TEMP. 6E. 400.0 AND LE. 1800.0 R
00101 8* C
00103 9* C
00112 10* C
00113 11* C DATA X1,X2,X3,X4 /16.55,-0.4933,-1.935,0.2283,/T0,DT /459.67,400./
00114 12* C THETA = (T-T0)/DT
00115 13* C ELAS = (((X4*THETA+X3)*THETA+X2)*THETA+X1)*1.0E06
00117 14* C IF (T.6E.400.0.AND.T.LE.1800.0) RETURN
00122 15* C WRITE (6,1) T
00122 16* C 1 FORMAT (1H0,57HMODULUS OF ELASTICITY OF COPPER, TEMP. OUT OF RANGE
00123 17* C RETURN
00124 18* C END
    
```

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* EMIT/SCZ93 \*\*\*\*\*

\*\*\*\*\* EMIT/SC293 \*\*\*\*\*  
QFOR,S ME\*NASAS.EMIT/SC293,ME\*NASAS.EMIT/SC293  
FOR S9A-07/13/72-20:56:16 (0,)

FUNCTION EMIT ENTRY POINT 000030

STORAGE USED: CODE(1) 000035; DATA(0) 000016; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK,\_NAME)

0003 POLY  
0004 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)  
0001 000016 1L 0000 R 000001 A 0000 R 000000 EMIT 0000 000011 INJPS 0003 R 000000 POLY

```

00101 1*          FUNCTION EMIT(T)
00101 2*          C
00101 3*          C THIS SUBROUTINE COMPUTES :
00101 4*          C THE TOTAL HEMISPHERICAL EXITTANCE AS A FUNCTION OF THE SURFACE
00101 5*          C TEMPERATURE (R)
00101 6*          C
00103 7*          DIMENSION A(5)
00104 8*          DATA A(1),A(2),A(3),A(4),A(5)/0.8990103E+00, -0.1400633E-03,
00104 9*          1 0.3879700E-06, -0.3937509E-09, 0.1015627E-12/
00112 10*         IF(T.6E+1600.0) GO TO 1
00114 11*         EMIT = POLY(5,A,T)
00115 12*         RETURN
00116 13*         1 EMIT = 0.228
00117 14*         RETURN
00120 15*         END

```

END OF COMPILATION: NO DIAGNOSTICS.

GH06/P \*\*\*\*\* ENTAIR \*\*\*\*\*

\*\*\*\*\* ENTAIR \*\*\*\*\*

QFOR,S ME\*NASAS5.ENTAIR,ME\*NASAS5.ENTAIR  
FOR S9A-07/13/72-20:56:20 (0,)

FUNCTION ENTAIR ENTRY POINT 000204

STORAGE USED: CODE(1) 000225; DATA(0) 000071; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 ALOG  
0004 SGR  
0005 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000157	IOL	0001	000103	SL	0000	R	000001	A	0000	R	000000	ENTAIR	0000	R	000032	ENTN		
0000	R	000033	ENTJ	0000	R	000013	ENTO	0000	R	000015	ENT00	0000	R	000007	FMN	0000	R	000006	FMO
0000	R	000053	INJ-S	0000	R	000031	TLOG	0000	R	000030	TMT	0000	R	000026	TMO	0000	R	000027	TOT
0000	R	000012	IT0	0000	R	000014	T00	0000	R	000022	T02	0000	R	000023	T03	0000	R	000024	T04
0000	R	000025	T05	0000	R	000016	T2	0000	R	000017	T3	0000	R	000020	T4	0000	R	000021	T5
0000	R	000011	WMN	0000	R	000010	WMO												

FUNCTION ENTAIR(T)

00101	1*	C
00101	2*	C
00101	3*	C
00101	4*	C
00101	5*	C
00101	6*	C
00103	7*	C
00104	8*	
00104	9*	
00112	10*	
00112	11*	
00121	12*	
00124	13*	
00125	14*	
00126	15*	
00127	16*	
00130	17*	
00131	18*	
00132	19*	
00133	20*	
00134	21*	
00135	22*	
00136	23*	
00137	24*	
00140	25*	
00142	26*	
00142	27*	
00143	28*	

THIS FUNCTION SUBPROGRAM COMPUTES :  
ENTHALPY OF AIR AS A FUNCTION OF TEMPERATURE (R)  
UNITS BTU/LBM

DIMENSION A(5)  
DATA A(1),A(2),A(3),A(4),A(5)/0.34240E+00,-0.95225E-03,  
0.31862E-05,-0.45750E-08,0.23750E-11/  
DATA FMO,FMN,WMO,W4MN,TT0,ENT0/0.234559,0.765441,31.9988,28.0134,  
1 600.,143.47 /  
DATA T00,ENT00/300.,71.61/  
T2 = T\*\*2  
T3 = T2\*T  
T4 = T2\*T2  
T5 = T2\*T3  
T02 = T00\*\*2  
T03 = T02\*T00  
T04 = T02\*T02  
T05 = T03\*T02  
TMO = T-T00  
TOT = T/TT0  
TMT = T-TT0  
TLOG = LOG(TOT)  
IF (T.GT. 600.) GO TO 5  
ENTAIR = ENT00+A(1)\*TMO+(A(2)/2.)\*(T2-T02)+(A(3)/3.)\*  
(T3-T03)+A(4)/4.)\*(T4-T04)+A(5)/5.)\*(T5-T05)  
1 RETURN

```

***** ENTAIR *****
00144 29*      5 ENTN = 9.47*TMT-3470.*TLOG-1.16E+06*(1./T-1./TTO)
00145 30*      ENTO = 11.515*TMT-344.*(SQRT(T)-SQRT(TTO))+1530.*TLOG
00146 31*      IF (T .LT. 5000.) GO TO 10
00150 32*      ENTO = ENTO+5.E-05*(T2-TTO**2)/2*-4000.*TMT
00151 33*      10 ENTAIR = ENTO+FM0*ENTO/#MO+FMN*ENTN/MIN
00152 34*      RETURN
00153 35*      END

```

END OF COMPILATION: NO DIAGNOSTICS.

END6,P \*\*\*\*\* EXITAV \*\*\*\*\*

\*\*\*\*\* EXITAV \*\*\*\*\*

QFOR>S ME\*NASAS.EXITAV,ME\*NASAS.EXITAV  
FOR S9A-07/13/72-20:56:23 (0,)

SUBROUTINE EXITAV ENTRY POINT 000076

STORAGE USED: CODE(1) 000110; DATA(0) 000030; BLANK COMMON(2) 000000

COMMON BLOCKS:

0003 GRD 003721  
0004 QIN 001610  
0005 ABSRST 000601

EXTERNAL REFERENCES (BLOCK, NAME)

0006 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000012	1116	0001	000021	1146	0005	R	000000	ALPHFN	0005	R	000226	ALPHMP	0003	R	003465	CSF					
0003	003464	DX21	0003	003472	DX21	0003	R	003463	DZMFN	0003	R	003410	EBFN	0003	R	003441	EBMP					
0005	R	000461	EMITFN	0005	R	000512	EMITMP	0003	R	003467	EXTIFN	0003	R	003471	EXTIMP	0003	R	003466	EXTISFN			
0003	R	003470	EXTSMP	0005	R	000517	EXTVCT	0000	I	000002	I	0000	I	000010	INJPS	0000	I	000003	J			
0000	I	000004	K	0003	003461	LCT	0003	003462	LIT	0003	003460	MCVRD	0004	001605	NSRD	0004	000764	QSTB	0004	000000	TRMTX	
0004	001604	NTM	0004	000454	QIFN	0004	001274	QITB	0003	003670	SSTT	0000	R	000001	ZMP	0003	003453	ZZZ				
0004	001606	QTO	0003	003473	SS	0003	003446	XX2	0000	R	000000	ZFN										
0004	001607	TX	0003	003446	XX2	0000	R	000000	ZFN													

00101 1\* SUBROUTINE EXITAV  
00101 2\* C  
00101 3\* C THIS SUBROUTINE COMPUTES :  
00101 4\* C THE EXITATION VECTOR NECESSARY FOR THE CALCULATION OF THE INCIDENT  
00101 5\* C NET RADIANT FLUX  
00101 6\* C  
00101 7\* C COMMON /GRD/ TRMTX(30,60),EBFN(5,5),EBMP(5),XX2(5),ZZ2(5),  
00103 8\* MCVRD,LCT,LIT,DZMFN,  
00103 9\* DX2,CSF,EXTISFN,EXTIFN,EXTSMP,EXTIMP,DX21,SS(5,5,5),  
00103 10\* ,SSIT(5,5)  
00104 11\* COMMON /QIN/TM(100), QSFN(100,2), QIFN(100,2), QSTB(100,2),  
00104 12\* QITB(100,2), NTM, NSRD, QTO, TX  
00105 13\* COMMON /ABSRST/ALPHFN(5,5,6),ALPHMP(5,31),EMITFN(5,5),EMITMP(5),  
00105 14\* EXTVCT(50)  
00106 15\* ZFN = EXTISFN+EXTIFN  
00107 16\* ZMP = EXTIMP+EXTIMP  
00110 17\* DO 16 I=1,5  
00113 18\* DO 15 J=1,5  
00116 19\* K = (I-1)\*5 + J  
00117 20\* 15 EXTVCT(K) = EMITFN(J,I)\*EBFN(J,I)+(1.0-ALPHFN(I,J,6))\*ZFN  
00121 21\* K = I+25  
00122 22\* 16 EXTVCT(K) = EMITMP(I)\*EBMP(I)+(1.0-ALPHMP(I,31))\*ZMP

\*\*\*\*\* EXITAV \*\*\*\*\*

00124 23\* RETURN  
00125 24\* END

END OF COMPILATION: NO DIAGNOSTICS.

END,P \*\*\*\*\* FINI \*\*\*\*\*

\*\*\*\*\* FINT \*\*\*\*\*

QFOR,S ME\*NASAS.FINT,ME\*NASAS.FINT  
FOR S9A-07/13/72-20:56:25 (0,)

SUBROUTINE FINT ENTRY POINT 000075

STORAGE USED: CODE(1) 000111; DATA(0) 000027; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000043 1116 0000 R 000000 DDX 0000 000010 INJPS 0000 I 000001 J

```

00101 1* SUBROUTINE FINT(Y, Y0, DX, N, F)
00101 2* C
00101 3* C THIS SUBROUTINE COMPUTES :
00101 4* C THE INDEFINITE INTEGRAL BY SIMPSONS RULE OF THE EQUALLY-SPACED
00101 5* C ARRAY Y(N)
00101 6* C
00103 7* DIMENSION Y(N), F(N)
00104 8* F(1) = Y0
00105 9* DDX = DX/24.
00106 10* F(2) = F(1)+DDX*(9.*Y(1)+19.*Y(2)-5.*Y(3)+Y(4))
00107 11* F(3) = F(2)+DDX*(-Y(1)+13.*Y(2)+Y(3))-Y(4)
00110 12* DO 1 J=4,N
00113 13* 1 F(J) = F(J-1)+DDX*(9.*Y(J)+19.*Y(J-1)-5.*Y(J-2)+Y(J-3))
00115 14* RETURN
00116 15* END
    
```

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* FLSTRT \*\*\*\*\*

\*\*\*\*\* FLSTAT \*\*\*\*\*  
 @FOR'S ME\*NASAS.FLSTRT\*ME\*NASAS.FLSTRT  
 FOR S9A-07/13/72-20:56:28 (0.)

SUBROUTINE FLSTRT ENTRY POINT 000231

STORAGE USED: CODE(1) 000245; DATA(0) 000233; BLANK COMMON(2) 001115

COMMON BLOCKS:

0003 DVCWFL 000002  
 0004 FLCMP 000024

EXTERNAL REFERENCES (BLOCK, NAME)

0005 DERIVL  
 0006 CNTLN  
 0007 RHOF  
 0010 CPF  
 0011 TRNSPT  
 0012 DDX  
 0013 RKSF  
 0014 NERR2\$  
 0015 NWDJ\$  
 0016 NIO2\$  
 0017 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000046	10L	0000	000051	115F	0000	000127	120F	0001	000076	15L	0001	000120	20L												
0001	000143	25L	0001	000010	5L	0000	R	000006	AL	0002	000562	AUX2	0006	R	000000	CNTLN										
0002	001070	CP0	0002	001107	COH	0002	000574	CONFN	0002	000740	CONMP	0010	R	000000	CPF											
0004	R	000000	DI-META	0002	001072	DELTA	0000	R	000014	DELY	0005	R	000000	DERIVL												
0000	R	000030	DYST	0002	001110	DXI	0002	001053	DXIMP	0002	001052	DXITB	0000	R	000000	D0B										
0003	000001	FLUXI	0002	R	001051	OZ	0002	001103	FENTH	0002	001045	FFO	0002	001101	FLUX											
0002	R	001042	FPR	0002	R	001037	FR	0002	R	001040	FNU	0002	001046	FOF	0002	000012	FP									
0002	001104	FREJ	0002	001043	FRL	0002	001044	FRM	0002	001111	FRD	0002	001036	FRE	0000	R	000047	FT1								
0002	000024	FW	0002	001113	FXHW	0002	001112	FXOH	0002	R	001047	FZ	0002	001105	HFN	0002	001105	HFN								
0000	I	000043	IBKP	0000	I	000045	IERR	0002	I	001003	IFLOW	0000	000223	INJPS	0000	I	000042	ISTP								
0002	001024	LC	0002	001032	LIM	0002	001031	LIMWRT	0002	001011	LL1	0002	001012	LL2	0002	001012	LL2	0002	001012	LL2						
0002	001006	LMP	0002	001014	LL4	0002	001015	LL5	0002	001016	LL6	0002	001017	LL7	0002	001017	LL7	0002	001017	LL7						
0002	001030	MM1	0002	001004	LT	0002	001005	LTB	0002	001007	LTBMZ	0002	001010	LTB2MZ	0002	001010	LTB2MZ	0002	001010	LTB2MZ						
0002	001035	NCONV	0002	I	001002	NCTL	0002	001025	MSTOTR	0002	I	000777	MZ	0002	001034	NCCZ	0002	001034	NCCZ	0002	001034	NCCZ				
0002	001027	NW1	0002	001001	NRMP	0002	001026	NCTM	0002	I	000046	NEG	0002	001023	NEGUS	0002	001023	NEGUS	0002	001023	NEGUS					
0002	001021	NRTB1	0000	I	000044	NTRY	0002	001020	NRMP1	0002	001022	NRMP2	0002	001000	NRTB	0002	001000	NRTB	0002	001000	NRTB					
0002	001062	PHIM	0002	001071	PI	0002	000776	NX	0002	R	000017	PD	0002	001063	PHIF	0002	001063	PHIF	0002	001063	PHIF					
0002	000524	GRMP	0002	001074	ROTWRT	0002	R	000000	RHOF	0002	001114	RHOFN	0002	000360	GRFN	0002	000360	GRFN	0002	000360	GRFN					
0000	R	000011	RL	0002	001073	RLIMIT	0002	001100	RHEND	0000	R	000022	SD	0002	001067	RH00	0002	001067	RH00	0002	001067	RH00				
0002	R	001077	TI	0002	001076	TINTL	0002	000276	TMP	0002	000536	TREF	0002	000050	TW	0002	000050	TW	0002	000050	TW					
0002	R	001064	T0	0003	R	000000	WRAT	0002	R	001066	W0	0002	000536	XIFN	0002	000005	XIMP	0002	000005	XIMP	0002	000005	XIMP			
0002	000000	XITB	0002	001106	XL	0002	000050	XP	0002	000050	XX	0002	001054	XRE	0002	001057	XX10	0002	001057	XX10	0002	001057	XX10			
0002	001060	XX11	0002	001061	XX12	0002	001055	XX3	0002	001055	XX4	0002	001056	XX4	0000	R	000000	Y	0000	R	000000	Y	0000	R	000000	Y



0000 R 000025 YS

0000 R 000036 YSIMP

0000 R 000033 YST

0000 R 000041 Z

0002 000550 ZETA

```

00101 1* SUBROUTINE FLSTAT(RE,PRL,DLTA)
00101 2*
00101 3* THIS SUBROUTINE PROVIDES :
00101 4* 1- FLUID FLOW (INITIAL AND CURRENT) CONDITIONS
00101 5* 2- TABLE HEADING FOR INITIAL CONDITIONS
00101 6* C
00101 7*
00103 COMMON XITB(5),XIMP(5),FP(10),FW(10),FT(10),TW(10,S),TEMP(10,10),
00103 TMP(10,5),QRFN(10,10),QRP(10),XIFN(10),ZETA(10),AUX2(10),
00103 CONFN(10,10),CONMP(10),COB(10),DOB(10)
00104 COMMON NX,MZ,NRTB,NRMP,NCTL,IFLOW,LT,LTB,LMP,LTBMZ,LTB2MZ,LL1,
00104 LL2,LL3,LL4,LL5,LL6,LL7,NRMP1,NRTB1,NRMP2,NEGUS,LC,MSTOTR,
00104 NCTM,NM1,MMLI,MRTI,LIV,MOD,NCCZ,NCONV
00105 COMMON FRE,FR,FNU,FMACH,FPR,FRL,FRM,FFO,FOF,FZ,FRAD,DZ,DXITB,DXIMP
00105 XRE,XX3,XX4,XX10,XX11,XX12,PHIM,PHIF,T0,P0,W0,RH00,CP0,PI
00105 DELTA,RLIMIT,ROTWRT,TREF,TINTL,TI,RTEND,FLUX,QREF,FENTH,
00105 FREJ,HFN,XL,COH,DXI,FRD,FXOH,FXHW,RHOFN
00106 COMMON /DVCWFL/ WRAT,FLUXI
00106 COMMON /FLCWP/ OTHETA(20)
00107 19*
00110 20* DIMENSION Y(3),DY(3),AL(3),RL(3),DELY(3),PD(3),SD(3),YS(3),
00110 DYST(3),YST(3),YSIMP(3)
00111 21* :
00111 22* C
00111 23*
00112 24* GO TO (5,10),IFLOW
00113 25*
00114 26* RH00 = RHOF(P0,T0)
00115 27* CP0 = CPF(RH00,T0)
00116 28* FZ = P0/(RH00*T0*CP0*778.16)
00117 29* FMACH = W0**2/(T0*CP0*778.16)
00120 30*
00120 31* 10 CALL TRNSPT (RE,PRL,DLTA,FR,FNU)
00120 32* C
00121 33* C
00122 34* INITIALIZATION
00123 35* Z = 0.0
00124 36* ISTOP = 1
00125 37* NTRY = 1
00126 38* IERR = 0
00127 39* NCTL = 1
00130 40* Y(1) = 1.0
00131 41* GO TO (15,20),IFLOW
00132 42* 15 Y(2) = RH00/RHOF(P0,TI)*WRAT
00133 43* Y(3) = FT(1)
00134 44* NEG = 3
00136 45* WRITE (6,115)
00137 46* GO TO 25
00140 47* 20 FT1 = FT(1)*T0
00141 48* Y(2) = RH00/RHOF(P0,FT1)*WRAT
00142 49* NEG = 2
00143 50* CALL DOX(FT,DTHETA,DZ,MZ)
00143 51* CALL RKSF (DERIVL,CNTLN,Y,DY,AL,RL,Z,DZ,NEG,ISTP,IBKP,NTRY,
00144 52* IERR,DELY,P0,SD,YS,YST,DYST,YSIMP)
00144 52* IF (IFLOW.EQ.2) RETURN

```

\*\*\*\*\* FLSTRT \*\*\*\*\*

```

00146 53* XP = FPR/2.0
00147 54* WRITE (6,120) P0,W0,T0,RE,PRL,DLTA,XP
00160 55* IFLOW = 2
00161 56* RETURN
00162 57* 115 FORMAT(1H,15X,23HINITIAL LINE CONDITIONS,,16X,23H***** **
00162 58* 1*****,,25X,31H(ALL QUANTITIES ARE NORMALIZED),,83X,5HFLUID,15
00162 59* 2X,4HWALL,,15X,16HPT.NO. POSITION,11X,8HPRESSURE,12X,8HVELOCITY,
00162 60* 31X,11HTEMPERATURE,9X,11HTEMPERATURE,,26X,1HZ,18X,1HP,19X,1HW,19X
00162 61* 4,1HT,18X,3HTWI,/)
00163 62* 120 FORMAT(1H0,5X,25HINLET PRESSURE P0 = ,F10,3,11H LBF/SQ.FT,/,
00163 63* 6X,25HREF. VELOCITY W0 = ,F10,5,11H FT/SEC ,/,
00163 64* 6X,25HREF. TEMPERATURE T00 = ,F10,3,11H R ,/,
00163 65* 6X,25H REYNOLDS NO = ,E20,5,/,
00163 66* 6X,25H PRANDTL NO = ,F20,6,/,
00163 67* 6X,25H DELTA = ,F20,6,/,
00163 68* 6X,25H REL.PRESSURE IS ,E20,6,/)
00163 69* C
00164 70* END

```

END OF COMPILATION: NO DIAGNOSTICS.

GDG:P \*\*\*\*\* FMINV \*\*\*\*\*

9FOR,S ME\*NASAS.FMINV,ME\*NASAS.FMINV  
FOR 59A-07/13/72-20:56:31 (0,)

SUBROUTINE FMINV ENTRY POINT 000254

STORAGE USED: CODE(1) 000306; DATA(0) 001250; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR33

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000046	105G	0001	000151	11L	0001	000051	111G	0001	000071	117G	0001	000104	123G	
0001	000125	133G	0001	000130	137G	0001	000172	150G	0001	000175	154G	0001	000227	164G	
0001	000212	20L	0000	R	001214	AA	0000	R	001217	B	0000	I	001224	INJPs	
0000	I	001215	I1	0000	I	001220	I2	0000	I	001213	J	0000	R	000000	XMAT

```

00101 1* SUBROUTINE FMINV (A,X,N,M)
00103 2* DIMENSION A(N,N),X(N),XMAT(25,26)
00104 3* DO 1 I=1,N
00107 4* XMAT(I,M) = X(I)
00110 5* DO 1 J=1,N
00113 6* XMAT(I,J) = A(I,J)
00116 7* DO 20 I=1,N
00121 8* AA = XMAT(I,I)
00122 9* DO 5 J=1,M
00125 10* XMAT(I,J) = XMAT(I,J)/AA
00127 11* IF (I.EQ.1) GO TO 11
00131 12* I1 = I-1
00132 13* DO 10 K=1,I1
00135 14* B = XMAT(K,I)
00136 15* DO 10 J=1,M
00141 16* 10 XMAT(K,J) = XMAT(K,J) - XMAT(I,J) * B
00144 17* 11 I2 = I+1
00146 18* IF (I.EQ.N) GO TO 20
00147 19* DO 15 K=I2,N
00152 20* B = XMAT(K,I)
00153 21* DO 15 J=1,M
00156 22* 15 XMAT(K,J) = XMAT(K,J) - XMAT(I,J) * B
00161 23* 20 CONTINUE
00163 24* DO 25 I=1,N
00166 25* 25 X(I) = XMAT(I,M)
00170 26* RETURN
00171 27* END

```

END OF COMPILATION: NO DIAGNOSTICS.

\*\*\*\*\* FMINV \*\*\*\*\*

BHDG:P \*\*\*\*\* HFL/CFFC43 \*\*\*\*\*

DATE 071372

PAGE 2

9FOR5 ME\*NASAS5.HFL/CFFC43\*ME\*NASAS5.HFL/CFFC43  
FOR S9A-07/13/72-20:56:34 (0P)

FUNCTION HFL ENTRY POINT 00022

STORAGE USED: CODE(1) 000024; DATA(0) 000014; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000000 HFL 0000 000007 INJPS 0000 R 000004 THETA 0000 R 000003 T0 0000 R 000001 X1  
0000 R 000002 X2

00101	1*	FUNCTION HFL(RHO,T)
00101	2*	ENTHALPY AS A FUNCTION OF DENSITY (SLUG/CU.FT) AND
00101	3*	TEMPERATURE (R) OF FC-43
00101	4*	UNITS BTU/SLUG
00103	5*	DATA X1,X2 /-0.020092,5.4054E-04/,T0 /401.677
00107	6*	THETA = T-T0
00110	7*	HFL = (X1+X2*(T+T0)/2.0)*THETA*32.174
00111	8*	RETURN
00112	9*	END

END OF COMPILATION: NO DIAGNOSTICS.

9H06,P \*\*\*\*\* HFL/CFFC75 \*\*\*\*\*

\*\*\*\*\* HFL/CFFC75 \*\*\*\*\*  
GFOR15 ME\*NASAS.HFL/CFFC75\*ME\*NASAS.HFL/CFFC75  
FOR S9A-07/13/72-20:56:36 (0,)

FUNCTION HFL ENTRY POINT 000022

STORAGE USED: CODE(1) 000024; DATA(0) 000014; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000000 HFL 0000 000007 INJPS 0000 R 000004 THETA 0000 R 000003 T0 0000 R 000001 X1  
0000 R 000002 X2

00101 1\* FUNCTION HFL(RHO,T)  
00101 2\* C  
00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES:  
00101 4\* C ENTHALPY AS A FUNCTION OF DENSITY (SLUG/CU.FT) AND  
00101 5\* C TEMPERATURE (R) OF FC-75  
00101 6\* C UNITS BTU/SLUG  
00101 7\* C  
00101 8\* DATA X1,X2 /0.115082,2.4333E-04/,T0 /324.67/  
00107 9\* THETA = T-T0  
00110 10\* HFL = (X1+X2\*(T+T0)/2.0)\*THETA\*32.174  
00111 11\* RETURN  
00112 12\* END

END OF COMPILATION: NO DIAGNOSTICS.

9HDG,P \*\*\*\*\* HFL/CFHE \*\*\*\*\*

\*\*\*\*\* HFL/CFHE \*\*\*\*\*

FOR S ME\*NASAS.HFL/CFHE,ME\*NASAS.HFL/CFHE  
FOR S9A-07/13/72-20:56:37 (0,)

FUNCTION HFL ENTRY POINT 000110

STORAGE USED: CODE(1) 000124; DATA(0) 000047; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 PF  
0004 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000005 A	0000 R 000002 A1	0000 R 000003 B1	0000 R 000004 C
0000 R 000017 DRH02	0000 R 000015 DT	0000 R 000015 DTM2	0000 R 000000 HFL
0000 R 000022 P	0003 R 000000 PF	0000 R 000001 R	0000 R 000010 RH00
0000 R 000014 TM2	0000 R 000012 TX	0000 R 000007 T0	0000 R 000021 UFL
0000 R 000020 Y			0000 R 000006 U0

00101 1\* FUNCTION HFL(RH0,T)

00101 2\* C  
 00101 3\* C  
 00101 4\* C  
 00101 5\* C  
 00101 6\* C  
 00101 7\* C  
 00103 8\* C  
 00103 9\* C  
 00114 10\* C  
 00115 11\* C  
 00116 12\* C  
 00117 13\* C  
 00120 14\* C  
 00121 15\* C  
 00122 16\* C  
 00123 17\* C  
 00124 18\* C  
 00125 20\* C  
 00126 21\* C  
 00127 22\* C  
 00130 23\* C

THIS FUNCTION SUBPROGRAM COMPUTES :  
 ENTHALPY AS A FUNCTION OF DENSITY (SLUG/CU.FT) AND  
 TEMPERATURE (R) OF HELIUM  
 UNITS BTU/SLUG

DATA R,A1,B1,C,A /2077.02,136.9595,3.5002295E-03,10.000658,1.49610  
 13E-02/,U0,T0,RH00 /39922.0,10.938889,4.669193/  
 RH0X = RH0\*515.4275  
 TX = T/1.8  
 DT = TX-T0  
 TM2 = 1.0/(TX\*TX)  
 DTM2 = 1.0/(T0\*T0)-TM2  
 DRH0 = RH00-RH0X  
 DRH02 = RH0X\*RH0X-RH00\*RH00  
 Y = 3.0\*R\*C\*TM2  
 UFL = U0+R\*(1.5\*DT+141.22973\*DTM2)+(Y+A1)\*DRH0-(Y\*B1-A1\*A)\*DRH02/  
 12.0  
 P = PF(RH0,T)  
 HFL = UFL\*0.013844+P/(RH0\*778.26)  
 RETURN  
 END

END OF COMPILATION: NO DIAGNOSTICS.

GM0G:P \*\*\*\*\* HFL/CFNAK. \*\*\*\*\*

\*\*\*\*\* HFL/CFNAK \*\*\*\*\*

QFOR S ME\*NASAS.HFL/CFNAK,ME\*NASAS.HFL/CFNAK  
FOR S9A-07/13/72-20:56:40 (0,)

FUNCTION HFL ENTRY POINT 000100

STORAGE USED: CODE(1) 000112; DATA(0) 000037; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 CAPPA  
0004 PF  
0005 ALOG  
0006 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000017 C	0003 R 000000 CAPPA	0000 R 000010 C1	0000 R 000011 C2	0000 R 000007 DT
0000 R 000000 HFL	0000 R 000014 H1	0000 R 000020 H2	0000 000030 INJPS	0004 R 000000 PF
0000 R 000012 P0	0000 R 000016 P1	0000 R 000013 THETA	0000 R 000015 TK	0000 R 000006 T0
0000 R 000001 X1	0000 R 000002 X2	0000 R 000003 X3	0000 R 000004 X4	0000 R 000005 X5

FUNCTION HFL(RHO,T)

00101 1\*  
00101 2\* C  
00101 3\* C  
00101 4\* C  
00101 5\* C  
00101 6\* C  
00101 7\* C  
00103 8\*  
00103 9\*  
00116 10\*  
00117 11\*  
00117 12\*  
00120 13\*  
00121 14\*  
00122 15\*  
00123 16\*  
00124 17\*  
00125 18\*  
00126 19\*

THIS FUNCTION SUBPROGRAM COMPUTES :  
ENTHALPY AS A FUNCTION OF DENSITY (SLUG/CU.FT) AND  
TEMPERATURE (R) OF NAK 78.6  
UNITS BTU/SLUG

DATA X1,X2,X3,X4,X5 /0.2255,-0.016292,0.005396,-0.000758,0.000054/  
1,T0,DT /469.67,300.0/C1,C2 /58.773064,-0.008433/P0 /2116.224/  
THETA = (T-T0)/DT  
H1 = (((X5\*THETA/5.0+X4/4.0)\*THETA+X3/3.0)\*THETA+X2/2.0)\*THETA  
1+X1)\*(T-T0)  
TK = CAPPA(RHO,T)  
P1 = PF(RHO,T)-P0  
C = C1+C2\*T  
H2 = ((1.0+T\*C2/C)\*ALOG(1.0+TK\*P1))/(C\*TK\*778.26)  
HFL = (H1+H2)\*32.174  
RETURN  
END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG:P \*\*\*\*\* HFL/CFSIL \*\*\*\*\*





```

***** HFL/CFSIL *****
00145      P1 = P/144.0-14.696
00146      Z(1) = 0.0
00147      Z(2) = 1+T+C11/C
00150      Z(3) = (A11*T-A*Z(2))/2.0
00151      Z(4) = (T*(B11-A*A11)-B*Z(2))/6.0
00152      Z(5) = -(A*B11+A11*B)*T/8.0
00153      Z(6) = -B*B11*T/20.0
00154      H2 = POLY(6,Z,P1)
00155      HFL = (H1+H2/(C*337.37))*32.174
00156      IF (T.LT.360.67.OR.T.GT.860.67.OR.P.GT.146116.224.OR.P.GT.110116.2
00156      124.AND.T.LT.460.67) WRITE(6,1) T,P
00163      1 FORMAT (1H0,43HENTHALPY OF SILICON OIL, OUT OF RANGE, T = ,F10.5,6
00163      1H, P = ,F15.5,/)
00164      RETURN
00165      END

```

END OF COMPILATION: NO DIAGNOSTICS.

OH06.P \*\*\*\*\* INTERP \*\*\*\*\*

\*\*\*\*\* INTERP \*\*\*\*\*

QFOR,S ME,NASAS5,INTERP,ME,NASAS5,INTERP  
FOR 59A-07/13/72-20:56:45 (0,)

SUBROUTINE INTERP ENTRY POINT 000171

STORAGE USED: CODE(1) 000226; DATA(0) 000063; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 YINT  
0004 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000037 105G 0001 000054 110G 0001 000062 115G 0001 000106 124G 0001 000122 127G  
0001 000130 134G 0000 I 000012 I 0000 000022 INJPS 0000 I 000014 J 0000 I 000013 K  
0000 R 000015 P 0003 R 000000 YINT 0000 R 000000 YY

```

00101 1* C
00101 2* C
00101 3* C
00101 4* C
00101 5* C
00101 6* C
00101 7* C
00101 8* C
00101 9* C
00101 10* C
00103 11* C
00103 12* C
00104 13* C
00107 14* C
00112 15* C
00114 16* C
00117 17* C
00120 18* C
00123 19* C
00126 20* C
00131 21* C
00133 22* C
00136 23* C
00137 24* C
00142 25* C
00143 26* C

SUBROUTINE INTERP(NX1,MZ1,XX1,ZZ1,YY1,NX2,MZ2,XX2,ZZ2,YY2)
THIS SUBROUTINE MAPS A TWO-DIMENSIONAL FUNCTION YY1 FROM ONE GRID
(XX1,ZZ1) ONTO ANOTHER TWO-DIMENSIONAL FUNCTION YY2 WITH ANOTHER GRID
(XX2,ZZ2)
NX1,MZ1 NUMBER OF NODAL POINTS OF THE ORIGINAL GRID (EACH LESS
THAN OR EQUAL TO 10)
NX2,MZ2 NUMBER OF NODAL POINTS OF THE NEW GRID (EACH LESS THAN
OR EQUAL TO 10)
DIMENSION XX1(NX1), ZZ1(MZ1), YY1(10,10), YY(10), XX2(NX2),
1 ZZ2(MZ2), YY2(10,10)
DO 2 I = 1,NX1
DO 1 K = 1,MZ1
1 YY(K) = YY1(K,I)
DO 2 J = 1,MZ2
P = ZZ2(J)
2 YY2(J,I) = YINT(ZZ1,YY,MZ1,3,P)
DO 4 J = 1,MZ2
DO 3 K = 1,NX1
3 YY(K) = YY2(J,K)
DO 4 I = 1,NX2
P = XX2(I)
4 YY2(J,I) = YINT(XX1,YY,NX1,3,P)
RETURN
END

```

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* MAIN \*\*\*\*\*

\*\*\*\*\* MAIN \*\*\*\*\*

QFOR: S ME\*NASAS5.MAIN; ME\*NASAS5.MAIN  
FOR S9A-07/13/72-20:56:47 (0,)

MAIN PROGRAM

STORAGE USED: CODE(1) 003774; DATA(0) 006145; BLANK COMMON(2) 001115

COMMON BLOCKS:

- 0003 ADBH 000352
- 0004 DVM 000002
- 0005 VELALT 001132
- 0006 CNV 000010
- 0007 GRD 003721
- 0010 QIN 001610
- 0011 TC 000042
- 0012 ABRST 000601
- 0013 SSF 000010
- 0014 GEOM 000020
- 0015 FLDINL 000457
- 0016 DVCNFL 000002

EXTERNAL REFERENCES (BLOCK NAME)

- 0017 DERIVM
- 0020 CNTLM
- 0021 TCALC
- 0022 CLOSE
- 0023 ADIABH
- 0024 QINCIO
- 0025 TK
- 0026 RHOF
- 0027 VISC
- 0030 CPF
- 0031 THCF
- 0032 HFL
- 0033 FLSTRT
- 0034 THCTB
- 0035 THCMP
- 0036 DEFINT
- 0037 SHAPEF
- 0040 RKS
- 0041 NINTRS
- 0042 NRNL\$
- 0043 NWDUS
- 0044 NI01\$
- 0045 NI02\$
- 0046 NROUS
- 0047 NRBUS
- 0050 NWBUS
- 0051 NWNL\$
- 0052 SORT
- 0053 ASIN
- 0054 NSTOPS

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000	005361	1055F	0001	002064	10766	0001	001320	1100L	0001	002075	1104G	0001	002104	11126
0001	002113	1120G	0001	002203	1140G	0001	002245	1151G	0000	005372	1155F	0001	002275	1163G
0001	000732	12L	0001	002730	1246G	0001	002731	1251G	0000	005403	1255F	0001	002735	1256G
0001	002742	1264G	0001	003113	1324G	0000	005406	1355F	0001	003360	1410G	0001	003363	1413G
0001	003431	1424G	0001	003434	1427G	0001	003477	1440G	0001	003506	1443G	0000	003530	1452G
0000	005425	1455F	0001	003537	1461G	0001	003662	1500G	0000	005167	1555F	0000	005436	1555F
0000	005454	16F	0000	005447	2F	0001	001136	200L	0001	000032	200L	0001	001360	2100L
0000	005455	22F	0001	000120	227G	0001	000160	237G	0001	001110	24L	0001	000161	242G
0001	000221	250G	0000	005207	255F	0001	000235	263G	0001	001167	28L	0000	005450	3F
0001	001217	30L	0001	000141	300L	0001	002761	312L	0000	005457	313F	0001	003107	314L
0001	000336	316G	0000	005476	320F	0000	005722	321F	0001	000364	325G	0001	000410	337G
0001	001472	34L	0001	003461	341L	0001	000426	346G	0000	005243	355F	0001	000444	355G
0001	000456	360G	0001	003656	364L	0001	000474	372G	0000	005452	4F	0001	000506	400G
0001	000524	407G	0001	000552	416G	0001	000564	424G	0001	000576	432G	0001	000610	440G
0000	005257	450F	0001	000644	455G	0001	000656	463G	0001	000670	472G	0001	000702	500G
0001	000171	500L	0001	000714	506G	0001	000726	514G	0001	000764	535G	0001	001012	543G
0001	001022	550G	0001	001030	554G	0000	005274	555F	0001	001036	560G	0001	001053	571G
0001	001070	600G	0000	005310	655F	0001	001211	665G	0001	001247	676G	0001	001255	703G
0001	001315	710G	0001	001336	722G	0001	001345	727G	0001	001416	740G	0000	005320	755F
0001	000214	8L	0001	000240	800L	0000	005332	855F	0000	005344	955F	0000	006017	9998F
0001	003770	9909L	0000	R 000621	A	0011	R 000003	ABEMF	0000	R 004355	AITB	0003	R 000001	AL
0000	R 004270	ALPHIT	0013	R 000002	ALPFN	0001	R 004256	ALPHA	0012	R 000000	ALPHFN	0012	R 000226	ALPHMP
0005	R 000310	ALTA	0005	R 000454	ALTR	0000	R 004265	AMAN	0000	R 004264	AN	0000	R 004300	ANGINC
0000	R 004263	AT	0000	R 004371	ATOT	0002	R 000562	AUX2	0000	R 004361	BIOTNO	0000	R 004350	BOL
0000	R 004257	BT	0020	R 000000	CNTLM	0002	R 000752	COB	0006	R 000007	COEFF	0002	R 001107	COH
0003	R 001161	COFL	0002	R 000574	CONFN	0003	R 000173	CONFNA	0002	R 000740	CONMP	0003	R 000147	CP
0030	R 000000	CPF	0002	R 001070	CP0	0007	R 003465	CSF	0013	R 000002	CTA	0003	R 000000	D
0036	R 000000	DEFINT	0002	R 001072	DELTA	0000	R 001441	DELY	0017	R 000000	DERIVM	0000	R 004325	DITB
0000	R 004241	DIABI	0002	R 000764	DOB	0000	R 004347	DTA	0003	R 000326	JTL	0000	R 004401	DTME
0000	R 004266	DIARTE	0006	R 000003	DX	0002	R 001110	DXI	0000	R 004344	DXIFN	0002	R 001053	DXIMP
0002	R 001052	DXITB	0007	R 003464	DXX2	0002	R 003472	DXX21	0000	R 000311	DX	0000	R 003101	DYST
0002	R 001051	DZ	0007	R 003463	DZMFN	0007	R 003410	ERFN	0007	R 003441	ERMP	0003	R 000244	EFF
0003	R 000205	EMIS	0012	R 000461	EMITFN	0012	R 000512	EMITMP	0007	R 003467	EMITFN	0007	R 003471	EXTIMP
0007	R 003466	EXTSFP	0007	R 003470	EXTSMP	0012	R 000517	EXTVCT	0002	R 001103	FENTH	0002	R 001045	FFO
0000	R 004500	FI	0000	R 004444	FINA	0015	R 000456	FLDINT	0000	R 004364	FLMASI	0014	R 000016	FLMASS
0000	R 004365	FLMW	0000	R 004316	FLU	0000	R 004317	FLUID	0000	R 004457	FLUIDA	0002	R 001101	FLUX
0016	R 000001	FLUXI	0002	R 001041	FMACH	0000	R 004314	FNM	0000	R 004365	FNMASI	0000	R 004362	FNMASS
0000	R 004315	FNITL	0002	R 001040	FNU	0002	R 001046	FOF	0002	R 000012	FP	0002	R 001042	FPR
0002	R 001037	FR	0002	R 001050	FRAD	0002	R 001111	FRD	0002	R 001036	FRE	0002	R 001104	FREJ
0002	R 001043	FRL	0002	R 001044	FRM	0002	R 000036	FT	0002	R 000024	FW	0002	R 001113	FXHW
0002	R 001112	FXOH	0002	R 001047	FZ	0000	R 004260	GAMMA	0000	R 004750	GTNPT	0003	R 000013	H
0003	R 000231	HAB	0000	R 004231	HDG	0032	R 000000	HFL	0000	R 004356	HFL0	0002	R 001105	HFN
0000	R 004246	HFII	0000	R 004301	I	0000	R 004404	IBKP	0011	R 000001	ICASE	0000	R 004405	IERR
0002	R 001003	IFLOW	0000	R 004403	IFVD	0000	R 004304	IHI	0000	R 004305	IH2	0006	R 000002	IOPTN
0000	R 004237	ISM	0000	R 004323	IT	0011	R 000000	ITAPE	0000	R 004276	IU	0000	R 004322	IUL
0000	R 004277	IW	0000	R 004324	II	0000	R 004302	J	0000	R 004311	K	0000	R 004307	K1
0000	R 004310	K2	0002	R 001024	LC	0007	R 003461	LCT	0015	R 000001	LFLD	0002	R 001032	LIM
0002	R 001031	LIMVRT	0002	R 001011	LL1	0002	R 001012	LL3	0002	R 001013	LL4	0002	R 001014	LL4
0002	R 001015	LL5	0002	R 001016	LL6	0002	R 001017	LL7	0002	R 001006	LMP	0002	R 001004	LT
0002	R 001005	LTB	0000	R 004376	LTBJ	0002	R 001007	LTBMZ	0002	R 001010	LTB2MZ	0000	R 004271	LTS
0007	R 003462	LTT	0003	R 000064	M	0000	R 004700	MANIFD	0007	R 003460	MCVRD	0002	R 001101	MDOT
0000	R 000000	MDOTI	0000	R 004345	MWEP	0002	R 001030	MW1	0002	R 001033	MODA	0000	R 004474	MRI
0000	R 004272	MRSTRT	0002	R 001025	MSTOTR	0002	R 000777	MZ	0005	R 001033	NA	0002	R 001034	NCCZ
0002	R 001035	NCONV	0002	R 001002	NCTL	0002	R 001026	NCTM	0002	R 001023	NEQUS	0015	R 000000	NFLDTA
0002	R 001027	NMI	0005	R 001131	NR	0002	R 001001	NRMP	0002	R 001020	NRMP1	0002	R 001022	NRMP2

\*\*\*\*\* MAIN \*\*\*\*\*

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0002 I 001000 NRTB
0010 I 001505 NSRD
0000 I 004402 NTRY
0000 R 004261 PHI
0002 R 001071 PI
0000 R 004320 PRO
0010 R 000454 QIFN
0002 R 000360 GRFN
0000 R 001131 R
0002 R 001114 RHOFN
0000 R 004251 RHOMPI
0002 R 001073 RLIMIT
0013 R 000000 R2
0000 R 004336 SWP
0007 R 003473 SS
0000 R 004330 STIP
0013 R 000001 T
0000 R 004312 TBW
0000 R 004267 TEND
0034 R 000000 THCTB
0003 R 000040 TIN
0015 R 000002 TMEFLD
0002 R 001075 TREF
0000 R 004426 TUBEA
0000 R 004253 VELM
0000 R 006020 W
0000 R 004337 WWC
0000 R 006020 W1
0002 R 000000 X1TB
0013 R 000004 XSPM
0007 R 003446 XX2
0000 R 000001 Y
0007 R 003453 ZZ2

0002 I 004244 NRTBI
0003 I 000076 NT
0000 I 004275 NT1
0002 R 001062 PHIF
0003 R 000052 PIN
0000 R 004255 PROB
0010 R 001274 QITB
0002 R 000524 GRMP
0002 R 001074 RDTWRT
0000 R 004250 RHOFNI
0000 R 004333 RHOTB
0000 R 004332 ROTB
0000 R 002261 SQ
0006 R 000005 SWPC
0007 R 003670 STT
0000 R 004247 STIPI
0005 R 000000 TA
0000 R 004366 TBMA51
0000 R 004373 TFLD
0000 R 004262 THETA
0002 R 001076 TINL
0002 R 002076 TMP
0000 R 004370 TOTMSI
0007 R 000000 TRMTX
0002 R 000050 TW
0005 R 000764 VELR
0015 R 000312 WIFLD
0000 R 004341 WWR1
0000 R 006033 W2
0002 R 001106 XL
0002 R 001050 XX10
0004 R 000000 XX20
0000 R 002571 Y5

0002 I 001021 NRTB1
0014 I 000012 NTRS
0002 I 000776 NX
0002 R 001062 PHIM
0000 R 004367 PLMASI
0000 R 004321 PROMTL
0000 R 000315 QMVL
0010 R 000144 QSFN
0000 R 004353 REY
0000 R 004252 RHOI
0000 R 004243 RHOTBI
0002 R 001100 RTEND
0014 R 000000 SFN
0000 R 004327 SROOT
0006 R 000000 STAGX
0014 R 000017 STR
0000 R 004254 TAU
0014 R 000014 TBMA55
0003 R 000026 TH
0002 R 001077 TI
0025 R 000000 TK
0000 R 004335 TNN
0003 R 004363 TOTM55
0003 R 000077 TSTAR
0010 R 001607 TX
0006 R 000001 VERTX
0016 R 000000 WRAT
0000 R 004341 WWR1
0000 R 004375 XFNU
0000 R 004360 XNUSLT
0002 R 001050 XX11
0002 R 001055 XX3
0000 R 003721 Y5IMP

0000 I 004306 NS
0000 I 004274 NTF
0000 R 001751 PD
0011 R 000004 PHIN
0014 R 000015 PLMA55
0000 R 004627 PROTLR
0003 R 000314 QOUT
0010 R 000764 QSTB
0026 R 000000 RHOF
0000 R 004252 RHOMET
0002 R 001067 RHOO
0000 R 004705 RUNOPT
0000 R 004273 SIDE
0000 R 004245 SROOTI
0000 R 004326 STB
0006 R 000006 ST4
0000 R 004357 TAVG
0000 R 004313 TBMTL
0031 R 000000 THCF
0015 R 000146 TIFLD
0003 R 000270 TL
0014 R 000013 TNXL
0003 R 000340 TOUT
0000 R 004303 TSTAR
0002 R 001064 T0
0027 R 000000 VISC
0013 R 000006 WMA
0000 R 004342 WWR2
0002 R 000536 XIFN
0002 R 001054 XRE
0002 R 001061 XX12
0000 R 001056 XX4
0000 R 000341 Y5T

0006 I 000004 NSRAD
0010 I 001604 NTM
0000 R 004372 PFLD
0000 R 004240 PHN
0000 R 004354 PRLNTL
0002 R 001102 GREF
0010 R 001606 QTO
0000 R 004374 RHOFLD
0000 R 004334 RHOMP
0000 R 004331 RITB
0013 R 000003 R1
0000 R 004351 SLOPE
0000 R 004340 SR02
0000 R 004242 STBI
0000 R 004410 SYSTEM
0003 R 000217 TB
0002 R 000132 TEMP
0035 R 000000 THCMP
0000 R 004400 TIME
0010 R 000000 TM
0011 R 000002 T0
0005 R 000144 TR
0000 R 004536 TUBE
0005 R 000620 VELA
0003 R 000135 VSC
0013 R 000007 WVB
0002 R 001066 W0
0002 R 000005 XIMP
0013 R 000005 XSPM
0000 R 004377 XX13
0004 R 000001 XX4B
0002 R 000550 ZETA

```

C THIS PROGRAM PROVIDES :  
 C 1- INPUT AND OUTPUT  
 C 2- UNIT CONVERSION  
 C 3- COMPUTATION OF DESIGN PARAMETERS  
 C 4- PREPARATION FOR PRINCIPLE INTEGRATION

INPUT DATA  
 \*\*\*\*\*

THE FOLLOWING CARDS DESCRIBE EACH VARIABLE IN THE NAMELISTS  
 WITH THEIR CORRESPONDING UNITS

NAMELIST /SYSTEM/  
 ISYM = 0 SINGLE, FLAT, SYMMETRICAL PANEL WITH STRAIGHT TUBES  
 = 1 CURVED PANELS, STRAIGHT TUBES  
 = 2 FLAT PANELS, U-SHAPED TUBES  
 = 3 CURVED PANELS, U-SHAPED TUBES

```

00100 1*
00100 2*
00100 3*
00100 4*
00100 5*
00100 6*
00100 7*
00100 8*
00100 9*
00100 10*
00100 11*
00100 12*
00100 13*
00100 14*
00100 15*
00100 16*
00100 17*
00100 18*
00100 19*

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***** MAIN *****
00100 20* C = 4 FLAT PANELS, STRAIGHT TUBES, NON-SYMMETRICAL LOADING
00100 21* C NUMBER OF RADIATING SIDES (1 OR 2)
00100 22* C NT NUMBER OF FLAT, SYMMETRICAL PANELS, EACH WITH NTBS TUBES
00100 23* C TSTAR ARRAY OF SINK TEMPERATURES FOR CARD INPUT
00100 24* C
00100 25* C NAMELISTS TUBEA, FINA, FLUIDA, AND MRI ARE USED ONLY FOR THE
00100 26* C NON-SYMMETRICAL CASE (ISYM>0)
00100 27* C
00100 28* C NAMELIST /TUBEA/
00100 29* C O INNER DIAMETER OF TUBE [IN]
00100 30* C AL TUBE LENGTHS FOR STRAIGHT TUBE CASE [FT]
00100 31* C W1 INLET-OUTLET TUBE SPACING FOR U-SHAPE CASE [IN]
00100 32* C W2 CROSSOVER TUBE SPACING FOR U-SHAPE CASE [IN]
00100 33* C
00100 34* C NAMELIST /FINA/
00100 35* C H HALF-WIDTHS OF FINS FOR STRAIGHT TUBE CASE [IN]
00100 36* C TH FIN THICKNESSES [IN]
00100 37* C ABEMF FIN ABSORBTIVITY/EMISSIVITY RATIO (USED WITH MRI INPUT)
00100 38* C
00100 39* C NAMELIST /FLUIDA/
00100 40* C TIN ENTRANCE (AND REFERENCE) TEMPERATURES [RJ]
00100 41* C PIN ENTRANCE (AND REFERENCE) PRESSURES [LBF/SQ.FT]
00100 42* C M COOLANT REFERENCE MASS FLOW RATE PER PANEL [LBM/HR]
00100 43* C
00100 44* C NAMELIST /MRI/
00100 45* C ITAPE UNIT NUMBER ATTACHED TO MRI TAPE
00100 46* C ICASE DATA CASE NUMBER ON MRI TAPE
00100 47* C NTM NUMBER OF DATA POINTS IN DATA CASE
00100 48* C TO TIME POINT IN ORBIT FROM WHICH TO TAKE DATA [MIN]
00100 49* C PHIN ANGLES OF PANELS RELATIVE TO MRI BASE DATA [DEGREES]
00100 50* C
00100 51* C THE NAMELIST QNML IS USED ONLY IN THE SYMMETRICAL CASE (ISYM=0)
00100 52* C
00100 53* C NAMELIST /QNML/
00100 54* C ITAPE UNIT NUMBER ATTACHED TO MRI TAPE
00100 55* C ICASE DATA CASE NUMBER ON MRI TAPE
00100 56* C NTM NUMBER OF DATA POINTS IN DATA CASE
00100 57* C TO STEADY-STATE TIME POINT AT WHICH TO TAKE DATA [MIN]
00100 58* C PHN ANGLE OF PANEL RELATIVE TO MRI BASE DATA [DEGREES]
00100 59* C
00100 60* C NAMELIST /TUBE/
00100 61* C DITBI TUBE INNER DIAMETER FOR SYMMETRICAL CASE [IN]
00100 62* C STBI TUBE THICKNESS [IN]
00100 63* C XL TUBE LENGTH FOR SYMMETRICAL CASE [FT]
00100 64* C RHOTBI TUBE DENSITY [LBM/CU.FT]
00100 65* C MZ NUMBER OF NODAL POINTS ALONG THE TUBE LENGTH (LESS THAN OR
00100 66* C EQUAL TO 10)
00100 67* C NRTBI NUMBER OF NODAL POINTS IN THE TUBE WALL RADIAL DIRECTION
00100 68* C (LESS THAN OR EQUAL TO 5)
00100 69* C NTBS NUMBER OF TUBES
00100 70* C
00100 71* C THE NAMELIST FLOW IS USED ONLY IN THE SYMMETRICAL CASE (ISYM=0)
00100 72* C
00100 73* C NAMELIST /FLOW/
00100 74* C MDOFI TOTAL COOLANT REFERENCE MASS FLOW RATE [LBM/HR]
00100 75* C TO ENTRANCE (AND REFERENCE) TEMPERATURE [RJ]
00100 76* C PO ENTRANCE (AND REFERENCE) PRESSURE [LBF/SQ.FT]

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\*\*\*\*\* MAIN \*\*\*\*\*

00100	77*	C	NAMLIST /FIN/ NX	NUMBER OF NODAL POINTS ALONG THE FIN HEIGHT (LESS THAN OR EQUAL TO 10)
00100	78*	C	SROOTI	FIN ROOT THICKNESS FOR SYMMETRICAL CASE [IN]
00100	79*	C	HFINI	FIN HEIGHT FOR SYMMETRICAL CASE [IN]
00100	80*	C	STIPI	FIN TIP THICKNESS FOR SYMMETRICAL CASE [IN]
00100	81*	C	RHOFNI	FIN DENSITY [LBM/CU.FT]
00100	82*	C	STAGX	DISTANCE FROM THE STAGNATION POINT ON THE SHUTTLE TO THE CENTER OF THE RADIATOR PANEL [FT]
00100	83*	C	VERTX	OVERALL DIMENSION OF THE RADIATOR PANEL IN THE DIRECTION PARALLEL TO THE ACCELERATION OF GRAVITY [FT]
00100	84*	C	NAMLIST /PROTLR/ NRMP	NUMBER OF NODAL POINTS IN THE PROTECTION LAYER RADIAL DIRECTION (LESS THAN OR EQUAL TO 5)
00100	85*	C	RHOMPI	PROTECTION LAYER DENSITY [LBM/CU.FT]
00100	86*	C	VELM	METEOROID VELOCITY [FT/SEC]
00100	87*	C	TAU	TIME THAT VULNERABLE AREA IS EXPOSED TO METEOROID ENVIRONMENT [DAYS]
00100	88*	C	PROB	PROBABILITY OF NO DAMAGE CAUSED BY METEOROID IMPACTS [DIMENSIONLESS]
00100	89*	C	ALPHA	EXPERIMENTAL CONSTANT THAT RELATES METEOROID FLUX AND MASS [G <sup>4</sup> /DAY.SQ.FT]
00100	90*	C	BTA	EXPERIMENTAL CONSTANT THAT RELATES METEOROID FLUX AND MASS [DIMENSIONLESS]
00100	91*	C	GAMMA	EMPIRICAL CONSTANT USED TO ADJUST PREDICTED PENETRATION DEPTH TO ONES OBSERVED EXPERIMENTALLY [DIMENSIONLESS]
00100	92*	C	PHI	EMPIRICAL CONSTANT [DIMENSIONLESS]
00100	93*	C	THETA	EMPIRICAL CONSTANT [DIMENSIONLESS]
00100	94*	C	ATK	EMPIRICAL CONSTANT USED TO ACCOUNT FOR SPALLING ON A TARGET OF FINITE THICKNESS [DIMENSIONLESS]
00100	95*	C	AN	EXPERIMENTAL CONSTANT THAT DESCRIBES PENETRATION DEPTH AS A FUNCTION OF ANGLE OF INCIDENCE [DIMENSIONLESS]
00100	96*	C	NAMLIST /MANIFD/ AMAN	TOTAL MANIFOLD AREA PROJECTED INTO THE FIN PLANE [SQ.FT]
00100	97*	C	NAMLIST /RUNOPT/ MSTOTR	= 1 TO COMPUTE STEADY STATE CONDITIONS = 2 TO SIMULATE TRANSIENT SYSTEM PERFORMANCE IN THE SYMMETRICAL CASE ONLY
00100	98*	C	DIWRTE	FIXED TIME INTERVAL BETWEEN DATA PRINTOUT DURING INTEGRATION [HR]
00100	99*	C	TEND	TERMINATION TIME FOR TRANSIENT PERFORMANCE CALCULATIONS [HR]
00100	100*	C	ALIMIT	ABSOLUTE ERROR LIMIT PER INTEGRATION STEP [DIMENSIONLESS]
00100	101*	C	RLIMIT	RELATIVE ERROR LIMIT PER INTEGRATION STEP [DIMENSIONLESS]
00100	102*	C	TI	INITIAL TEMPERATURE [R]
00100	103*	C	LIMWRT	MAXIMUM NUMBER OF DATA RECORDING DURING INTEGRATION TOWARD STEADY STATE RECORD
00100	104*	C	NCONV	= 0 NO AERODYNAMIC HEATING = 1 ASCENT = 2 REENTRY
00100	105*	C	LTT	= 0 OPTICAL PROPERTIES INDEPENDENT OF TEMPERATURE



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***** MAIN *****
00100 134* C
00100 135* C
00100 136* C
00100 137* C
00100 138* C
00100 139* C
00100 140* C
00100 141* C
00100 142* C
00100 143* C
00100 144* C
00100 145* C
00100 146* C
00100 147* C
00100 148* C
00100 149* C
00100 150* C
00100 151* C
00100 152* C
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00100 166* C
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00100 178* C
00100 179* C
00100 180* C
00100 181* C
00100 182* C
00100 183* C
00100 184* C
00100 185* C
00100 186* C
00100 187* C
00100 188* C
00100 189* C
00100 190* C

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= 1 OPTICAL PROPERTIES DEPENDENT ON TEMPERATURE
= 2 GRFN AND GRMP ARE BOTH EQUAL TO ZERO BUT QINCID IS CALLED
= 1 CONSTANT FLUID INLET CONDITIONS
= 2 VARIABLE FLUID INLET CONDITIONS
= 0 UNIFORM INITIAL TEMPERATURE EQUAL TO TI
= 1 INITIAL TEMPERATURE AS OBTAINED BY PREVIOUS COMPUTATIONS

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NAMELIST/GINPT/
IT CONTAINS THE VARIABLES IN MOST OF THE PREVIOUS NAMELISTS, AND IN ADDITION
HRSTR1 = 1 NEW VELOCITY AND ALTITUDE PROFILES
        = 2 NEW IRRADIATION HISTORY
        = 3 BOTH 1 AND 2
        = 4 NEW COOLANT INLET CONDITIONS
        = 5 BOTH 1 AND 4
        = 6 BOTH 2 AND 4
        = 7 ALL (1, 2, AND 4)
        = 8 ONLY NEW PARAMETERS FROM GINPT
IF THIS THIS NAMELIST IS OMITTED, THE PROGRAM WILL TERMINATE

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THE FOLLOWING CARDS DESCRIBE THE INPUT VARIABLES, OTHER THAN THOSE IN THE NAMELISTS, WITH THEIR CORRESPONDING UNITS

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NA  NUMBER OF ORDERED PAIRS OF ELEMENTS IN THE VELOCITY-TIME ARRAY FOR ASCENT PROFILES
NR  NUMBER OF ORDERED PAIRS OF ELEMENTS IN THE ALTITUDE-TIME ARRAY FOR REENTRY PROFILES
TA  ELEMENT OF TIME ARRAY (NA VALUES), SELECTED FOR ASCENT VELOCITY AND ELEVATION PROFILES [SEC]
TR  ELEMENT OF TIME ARRAY (NR VALUES), SELECTED FOR REENTRY VELOCITY AND ELEVATION PROFILES [SEC]
VELA ELEMENTS OF VELOCITY ARRAY (NA VALUES), SELECTED FOR ASCENT VELOCITY PROFILE OF ORBITER [FT/SEC]
VELR ELEMENTS OF VELOCITY ARRAY (NR VALUES), SELECTED FOR REENTRY VELOCITY PROFILE OF ORBITER [FT/SEC]
ALTA ELEMENT OF ALTITUDE ARRAY (NA VALUES), SELECTED FOR ASCENT ELEVATION PROFILE OF ORBITER [FT]
ALTR ELEMENT OF ALTITUDE ARRAY (NR VALUES), SELECTED FOR REENTRY ELEVATION PROFILE OF ORBITER [FT]
TBM,TBMTL TUBE MATERIAL SPECIFICATION (UP TO 12 ALPHANUMERIC CHARACTERS)
FNM,FNMTL FIN MATERIAL SPECIFICATION (UP TO 12 ALPHANUMERIC CHARACTERS)
FLU*FLUID FLUID SPECIFICATION (UP TO 12 ALPHANUMERIC CHARACTERS)
PRO*PROMTL PROTECTION LAYER MATERIAL SPECIFICATION (UP TO 12 ALPHANUMERIC CHARACTERS)
TMEFLO TIME AT WHICH FLUID INLET CONDITIONS ARE SPECIFIED [HR]
TIFLD INSTANTANEOUS FLUID INLET TEMPERATURE [R]
WIFLD INSTANTANEOUS FLUID MASS FLOW RATE [LBM/HR]

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\*\*\*\*\* MAIN \*\*\*\*\*

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00140 00140 248* NAMELIST/PROTLR,NRMP,RHOMPI,RHOMET,VELM,TAU,PROB,ALPHA,BTA,GAMMA,
00140 00140 249* PHI,THETA,ATK,AN
00141 00141 250* NAMELIST/MANIFD/ AMAN
00142 00142 251* NAMELIST/RUNOPT/MSTOTR,DTWRTE,TEND,ALIMIT,RLIMIT,TLIMWRT,NCONV,
00143 00143 252* LTT,FLD,LTS
00143 00143 253* NAMELIST/GINPT/DITBI,STBI,XL,RHDTBI,MZ,NRBTBI,NBTS,MDOITI,TO,PO,NX,
00143 00143 254* SROOTTI,HFNII,STIPI,RHOFNI,NRMP,RHOMPI,RHOMET,VELM,TAU,
00143 00143 255* PROB,ALPHA,BTA,GAMMA,PHI,THETA,ATK,AN,AMAN,MSTOTR,
00143 00143 256* DTWRTE,TEND,ALIMIT,RLIMIT,TLIMWRT,NCONV,STAGX,VERTX,
00143 00143 257* LTT,MRSTRT,NSRD,ITAPE,ICASE,NTM,TO,PHN
00143 00143 258*
00143 00143 259* C
00144 00144 260* C
00144 00144 261* 155 FORMAT(IH1,/,59X,A6,' SURFACE',/,54X,I2,I1,X,A6,A2,'-TUBE PANELS',
00145 00145 262* /,56X,I1,' RADIATING ',A5,/)
00145 00145 263* :55 FORMAT(//, MRI TAPE PARAMETERS: ,7X,'UNIT',I3,4X,' CASE',
00145 00145 264* 1 13,3X,I4,' DATA POINTS',3X,F8.3,' MINUTES INTO ORBIT',//,
00146 00146 265* 2 , FIN ABSORBTIVITY/EMISSIVITY RATIO: ,F8.4)
00146 00146 266* 355 FORMAT(//, FIN ANGLES RELATIVE TO MRI TAPE DATA (DEGREES)),
00147 00147 267* 1 /,10(2X,1PE10.4))
00147 00147 268* :55 FORMAT(//, FIN ANGLES AT OUTLET TUBES (MRI RELATIVE DEGREES)),
00150 00150 269* 1 /,10(2X,1PE10.4))
00151 00151 270* :55 FORMAT(//, TUBE SPECIFICATIONS',6X,1PE10.4,
00152 00152 271* , INCHES INSIDE DIAMETER',)
00152 00152 272* 655 FORMAT(//, TUBE LENGTH (FEET),/,10(2X,1PE10.4))
00153 00153 273* 755 FORMAT(//, INLET-OUTLET TUBE SPACING (INCHES),/,11(2X,1PE10.4))
00154 00154 274* :55 FORMAT(//, CROSSOVER TUBE SPACING (INCHES),/,11(2X,1PE10.4))
00154 00154 275* 1 /,10(2X,1PE10.4))
00155 00155 276* 1:55 FORMAT(//, FIN HALF-WIDTH (INCHES),/,11(2X,1PE10.4))
00156 00156 277* 1:55 FORMAT(//, SINK TEMPERATURE (DEGREES R),/,10(2X,1PE10.4))
00157 00157 278* 1:55 FORMAT(10(2X,1PE10.4))
00160 00160 280* 1:55 FORMAT(//, FLUID SPECIFICATIONS',//,
00161 00161 281* , INLET TEMPERATURE (DEGREES R),/,10(2X,1PE10.4))
00162 00162 282* 1:55 FORMAT(//, INLET PRESSURE (LBF/50.FT.),/,10(2X,1PE10.4))
00162 00162 283* 1:55 FORMAT(//, MASS FLOW RATE (LBM/HR),/,10(2X,1PE10.4))
00163 00163 284* C
00163 00163 285* C
00167 00167 286* DATA PI/3,1415927/,SIDE/,SIDE/,HDG/, FLAT', 'CURVED', 'STRAIGHT',
00170 00170 287* 1 'U-SHAPED',/
00170 00170 288* C
00170 00170 289* MRSTRT = 15
00173 00173 290* READ(5,SYSTEM)
00174 00174 291* NSRD = MAX(MIN(NSRD,2-AND(ISYM,1)),1)
00175 00175 292* NSRAD = NSRD
00177 00177 293* IF (ISYM,NE.0) GO TO 2000
00200 00200 294* NT = 1
00200 00200 295* GO TO 8
00204 00204 296* READ(5,TUBEA)
00207 00207 297* READ(5,FINA)
00212 00212 298* READ(5,FLUIDA)
00215 00215 299* READ(5,MRI)
00216 00216 300* NTF = (NT-1)*AND(ISYM,1)+1
00221 00221 301* NT1 = NT+1
00221 00221 302* IF (ITAPE,LE.0) GO TO 300
00223 00223 303* IF (ISYM,NE.3) GO TO 200
00224 00224 304* IU = (NT+1)/2
IW = NT/2+1

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00534 362* DO 20 J = 1,NSRD
00537 363* IU = J+2
00540 364* K1 = J*3-2
00541 365* K2 = K1+2
00542 366* DO 14 I = 1,NTM
00545 367* READ(5,3) TM(1),(QSFN(K,1),K=K1,K2),(QIFN(K,1),K=K1,K2),
1 (QSTB(K,1),K=K1,K2)
00545 368* IF (J.EQ.2) READ(3) NS,TM(1),(QSFN(K,1),QIFN(K,1),QSTB(K,1),K=1,3)
00564 369* 14 WRITE(IU) NS,TM(1),(QSFN(K,1),QIFN(K,1),QSTB(K,1),K=1,6)
00577 370* 20 CALL CLOSE(IU+2,1)
00612 371*
00612 372* C
00612 373* C MATERIAL SPECIFICATIONS READ IN
00614 374* 24 IF (MRSTRT.NE.15) GO TO 28
00616 375* READ(5,16) TBM,TBMTL,FNM,FNMTL,FLU,FLUID,PRO,PROMTL
00616 376* C
00616 377* C SYSTEM SPECIFICATIONS READ IN
00616 378* C
00630 379* 15 READ(5,TUBE)
00633 380* IF (ISYM.EQ.0) READ(5,FLOW)
00637 381* READ(5,FIN)
00642 382* READ(5,PROTLR)
00645 383* READ(5,MANIFD)
00645 384* C
00645 385* C PROGRAM CONTROL PARAMETERS READ IN
00645 386* C
00650 387* READ(5,RUNOPT)
00653 388* 16 FORMAT(2A6)
00653 389* C
00653 390* C FLUID INLET CONDITIONS READ IN
00653 391* C
00654 392* IF (LFLD.EQ.1) GO TO 30
00656 393* 28 IF (AND(MRSTRT,4).EQ.0) GO TO 30
00660 394* READ(5,2) NFLDTA
00663 395* READ(5,22) (TMEFLD(I),TIFLD(I),WIFLD(I),I=1,NFLDTA)
00673 396* 22 FORMAT (3F20.6)
00673 397* C
00674 398* 30 IUL = AND(ISYM,2)+1
00675 399* DO 840 I = 1,NT
00700 400* 840 M(I) = M(I)/(32.174*NTBS)
00702 401* DO 9997 IU = 1,IUL
00705 402* IF (AND(ISYM,2).EQ.0) GO TO 2100
00707 403* IW = AND(IU,1)
00710 404* H(I) = W(1,2-IW)
00711 405* H(NT+1) = MIN(W1(NT+1),W2(NT+1))
00712 406* IF (NT.EQ.1) GO TO 1100
00714 407* DO 1000 IT = 2,NT
00717 408* H(IT) = W(1,2-IW)/2
00721 409* 1000 DO 1300 IT = 1,NT
00724 410* AL(IT) = 0
00725 411* I1 = IT+1
00726 412* DO 1200 I = I1,NT1
00731 413* 1200 AL(IT) = W(1,IW)+AL(IT)
00733 414* 1300 AL(IT) = (2-AND(IU,1))*AL(IT)/12
00735 415* 2100 IF (ISYM.GT.0) CALL ADIABH(BOOL(AND(ISYM,3).EQ.3)*MOD(4-IU,3)+1)
00737 416* DO 9997 IT = 1,NT
00742 417* IF (ISYM.EQ.0) GO TO 34
00744 418* M5YOTR = 1

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***** MAIN *****
00745 419* IF (ITAPE.GT.0) PHN = PHIN(IT,IU)
00747 420* DITBI = D
00750 421* XL = AL(IT)
00751 422* MDOIT = M(IT)*32.174*NTBS
00752 423* T0 = TIN(IT)
00753 424* TIN(IT) = TOUT(IT)
00754 425* P3 = PIN(IT)
00755 426* SROOTI = TH(IT)
00756 427* STIPI = TH(IT)
00757 428* HFNI = (HAB(IT)*EFF(IT,1)+(2*H(IT+1)-HAB(IT+1))*EFF(IT,2))/
00757 429* ((EFF(IT,1)+EFF(IT,2))-D/2-STBI)
00760 430* 1 IF (AND(WRSTRT,2).NE.0) CALL GINCID (MSTOTR,PHN)
00760 431* 34 LISTING OF SYSTEM SPECIFICATIONS AND CONTROL PARAMETERS
00760 432* C
00762 433* WRITE(6,9998)
00764 434* WRITE(6,ONML)
00767 435* WRITE(6,TUBE)
00772 436* WRITE(6,FLOW)
00775 437* WRITE(6,FIN)
01000 438* WRITE(6,PROTLR)
01003 439* WRITE(6,MANIFD)
01006 440* WRITE(6,RUNOPT)
01006 441* C
01006 442* UYIT CONVERSION ON INPUT DATA
01006 443* C
01011 444* C
01012 445* DITB = DITBI/12.0
01013 446* STB = STBI /12.0
01014 447* SROOT = SROOTI/12.0
01015 448* STIP = STIPI/12.0
01016 449* HFNI = HFNI /12.0
01017 450* RIIB = DITB/2.0
01020 451* MDOIT = MDOIT/32.174
01021 452* RCTB = (RIIB+STB)*12.0
01022 453* RHOIB = RHOIBI/32.174
01023 454* RHOFN = RHOFNI/32.174
01024 455* RHOMP = RHOMPI/32.174
01024 456* C
01024 457* TNN = FLOAT(NTBS)
01024 458* C
01024 459* METEOROID PROTECTION LAYER THICKNESS
01025 460* SYPC = TK(GAMMA*ATK*BTA*RHOMET,THETA,PHI,AN,ALPHA,VELM,PROB,
01026 461* 1 SMP = TAU,RHOMPI,XL,TNN,AMPN,T0,ROTB)
01026 462* = SMP/12.0
01026 463* C
01026 464* INTERSECTION BETWEEN OUTER TUBE CIRCLE AND UPPER FIN PLANE
01026 465* C
01027 466* WWA = (STIP-SROOT)/(2.0*HFN)
01030 467* WWC = MWA**2+1.0
01031 468* SRO2 = SROOT/2.0
01032 469* WWR1 = RITH+STB
01033 470* WWR2 = WWR1+SMP
01034 471* WWB = SRO2-MWA*WWR1
01035 472* WWD = SQRT(WWC+WWR2**2-MWB**2)
01036 473* XSVP = ((-MWA*WWB+WWD)/WWC-WWR1)/HFN
01036 474* C
01036 475* CONTROL INTEGERS
01036 475* C

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01036 476* C
01037 477*
01040 478*
01041 479*
01042 480*
01043 481*
01044 482*
01045 483*
01046 484*
01047 485*
01050 486*
01051 487*
01052 488*
01053 489*
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01055 491*
01056 492*
01057 493*
01060 494*
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01067 502*
01070 503*
01071 504*
01072 505*
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01100 512*
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01124 523*
01125 524*
01126 525*
01126 526*
01130 527*
01131 528*
01132 529*
01133 530*
01134 531*
01136 532*

NRTB = NRTB1
NM1 = NX-1
MM1 = MZ-1
NRMP1 = NRMP-1
NRTB1 = NRTB-1
LT = MZ+NM1
LVP = LI+MZ
LVBZ = LTB+MZ
LTB2MZ = LTB+MZ*(NRTB-2)*MZ
LL1 = LL1-MZ
LL3 = LVP+MZ
LL4 = LL1+MZ
LL5 = LMP+NRMP2*MZ
LL6 = LL5-MZ
LL7 = LL6-MZ
LC = MZ/2
IF(INCONV.EQ.1) IOPTN = 0
IF(INCONV.EQ.2) IOPTN = 1

GRID INTERVALS
DZ = 1.0/FLOAT(MM1)
DXI = 1.0/FLOAT(NM1)
DXITB = STB/(RITB*FLOAT(NRTB1))
DXIFN = HPN/RITB*DXI
DXIMP = SWP/(RITB*FLOAT(NRMP1))
DELTA = DITB/XL

GRID SPACINGS
XITB(I) = 1.0
DO 50 I = 2,NRTB
XITB(I) = XITB(I-1)+DXITB
XIMP(I) = XIMP(I-1)+DXIMP
DO 55 I = 2,NRMP
ZETA(I) = 0.0
DO 60 J = 2,MZ
ZETA(J) = ZETA(J-1)+DZ
XIFN(I) = 0.0
DO 65 J = 2,NX
XIFN(J) = XIFN(J-1)+DXI
MMEP = (INT(2.0*XSMP/DXI)+1)/2
MCVRD = MMEP
IF(XSMP.GE.XIFN(MMEP+1)) MMEP = MMEP+1
DXX2 = (1.0-XSMP)/4.5
DXX21 = 2.0
XX2(I) = XSMP+DXX2/2.0
XSPM = (10.5+FLOAT(MCVRD))*DXI-XSMP/DXI
IF(MCVRD.EQ.0) XSPM = (DXI-2.0*XSMP)/DXI
ZZZ(I) = 0.0

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\*\*\*\*\* MAIN \*\*\*\*\*

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01137 533* DO 66 I = 2,5
01142 534* ZZZ(I) = ZZZ(I-1)+0.25
01143 535* XX2(I) = XX2(I-1)+DXX2
01143 536* C
01143 537* C
01143 538* C
01145 539* GENERAL CONSTANTS
01146 540* TOL = SROOT/(2.0*HFN)
01147 541* CTA = TOL-STIP/(2.0*HFN)
01150 542* DTA = SOROT(1.0+CTA**2)
01153 543* DO 70 I=2,NM1
01154 544* BOL = TOL-CTA*XIFN(I)
01155 545* OOB(I) = DTA/(2.0*BOL)
01157 546* C
01157 547* COB(I) = CTA/BOL
01160 548* SLOPE = (SROOT-STIP)/(2.0*HFN)
01161 549* SFN(I) = SROOT
01162 550* SFN(NX) = STIP
01165 551* DO 80 I = 2,NM1
01165 552* C
01165 553* C
01167 554* SFN(I) = STIP+2.0*HFN*SLOPE*(1.0-XIFN(I))
01170 555* C
01171 556* RH00 = RHOF(P0,T0)
01172 557* XRE = RHOF(P0,TI)
01173 558* FRL = 2.0*WDOI/(PI*TNN*RTB)
01174 559* FFO = (DELTA/2.0)**2
01175 560* FOF = 4.0/(XRE*DELTA)*RH00/RHOTB
01176 561* REY = FFO*RHOTB/RHOMP
01177 562* FRE = XRE/VISC(RH0I,TI)
01200 563* FRM = ((RTB+STB)/XL)**2
01201 564* FRD = 1.71E-09*T0**3
01202 565* FRD = FRD*RTB
01203 566* PRALNTL = (VISC(RH0I,TI)*CPF(RH0I,TI))/THCF(RH0I,TI)
01204 567* QREF = FRD*T0*TNN**2.0
01205 568* AITB = TNN*PI*RTB**2
01206 569* W0 = XRE/(DITB*3600.0*RH00)
01207 570* C20 = CPF(RH00,T0)
01210 571* HFL0 = HFL(RH00,T0)
01211 572* FENTH = FLUX*(HFL0+W0**2/1556.36)
01212 573* FREJ = PI*T0
01213 574* FXOH = (HFN/XL)**2
01214 575* FXHW = XL/(W0*3600.0*HFN**2)
01215 577* C
01216 578* T = SRO2/HFN
01217 579* R1 = WWR1/HFN
01220 580* R2 = WWR2/HFN
01220 581* DZMFN = XL/(HFN**4.0)
01221 582* C
01222 583* TREF = XL*DITB/XRE*RH00
01223 584* RDTWRT = DTWRT/TREF
01224 585* RTEND = TEND/TREF
01224 586* STR = T0*HFN*(SROOT*STIP)*TNN*SOROT(RH0FN/(PI*TREF))
01225 587* C
01226 588* XX3 = DXITB/DXIMP
01227 589* XX4 = DXITB/DXIFN
01227 589* XX10 = 1.0/(4.0*XIMP(NRMP)-DXIMP)

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\*\*\*\*\* MAIN \*\*\*\*\*

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01230 590* XX11 = 4.0*XX10*(2.0*XIMP(NRMP)-DXIMP)/(DXIMP**2)
01231 591* XX12 = 8.0*XX10*XIMP(NRMP)/DXIMP
01232 592* C
01233 593* PHIF = 4.0*ASIN(SROOT/(2.0*WRI+SMP))
01234 594* PHIM = 2.0*PI-4.0*ASIN(SROOT/(2.0*WRI+SMP))
01235 595* PHIF = PHIF*RTB/HFN
01236 596* COH = PHIM*WRI/(2.0*HFN)
01237 597* C
01238 598* COEFF = 3600./STAGX
01239 599* ST4 = 0.1714E-08*T0**4.0
01240 600* GTO = ST4
01241 601* DX = HFN*DXI*12.0
01242 602* C
01243 603* C
01244 604* C
01245 605* C
01246 606* INITIAL TEMPERATURE PROFILES
01247 607* IF (MRSTRT-LTS*8.LE.0) GO TO 312
01248 608* TINTL = TI/T0
01249 609* DO 310 J=1,NZ
01250 610* DO 302 I=1,NRTB
01251 611* T(I,J,I) = TINTL
01252 612* DO 305 I=1,NRMP
01253 613* TMP(J,I) = TINTL
01254 614* CONMP(J) = 0.0
01255 615* DO 310 I=1,NX
01256 616* CONFN(J,I) = 0.0
01257 617* TEMP(J,I) = TINTL
01258 618* C
01259 619* C
01260 620* C
01261 621* INITIAL FLOW CONDITIONS
01262 622* IFLOW = 1
01263 623* FT(1) = TW(1,1)
01264 624* IF (LFLO.EQ.2) FT(1) = TIFLD(1)/T0
01265 625* WRAT = 1.0
01266 626* FLUXI = MDOI
01267 627* IF (LFLO.EQ.2) WRAT = WIFLD(1)/FLUXI
01268 628* CALL FLSTRT(REY,PRNTL,DELTA)
01269 629* C
01270 630* C
01271 631* C
01272 632* TAVG = 0.5*(TI+T0)
01273 633* XNUSLT = FNU*DELTA*REY*PRNTL/4.0
01274 634* BIOTNO = XNUSLT/DITB*THCF(RH00,T0)*(STB/THCTB(TAVG)+SMP/THCMP
01275 635* (TAVG))
01276 636* 1 WRITE(6,313) XNUSLT,BIOTNO
01277 637* 1 FORMAT(1H,5X,25HINIT, NUSSELT NO. NU = ,E20.6//
01278 638* 6X,25HWALL BIOT NO. BI = ,E20.6//)
01279 639* IF(BIOTNO.GT.0.05) GO TO 314
01280 640* NRTB = 0
01281 641* LL4 = MZ*NX
01282 642* LL5 = LL4
01283 643* LL6 = MZ-2
01284 644* LTB = LL4
01285 645* NEGUS = MZ*(NX+1)
01286 646* C
01287 647* C
01288 648* C
01289 649* C
01290 650* C
01291 651* C
01292 652* C
01293 653* C
01294 654* C
01295 655* C
01296 656* C
01297 657* C
01298 658* C
01299 659* C
01300 660* C
01301 661* C
01302 662* C
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01306 666* C
01307 667* C
01308 668* C
01309 669* C
01310 670* C
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01332 692* C
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01340 700* C
01341 701* C
01342 702* C
01343 703* C
01344 704* C
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01346 706* C
01347 707* C
01348 708* C
01349 709* C
01350 710* C
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01364 724* C
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01367 727* C
01368 728* C
01369 729* C
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01371 731* C
01372 732* C
01373 733* C
01374 734* C
01375 735* C
01376 736* C
01377 737* C
01378 738* C
01379 739* C
01380 740* C
01381 741* C
01382 742* C
01383 743* C
01384 744* C
01385 745* C
01386 746* C
01387 747* C
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01389 749* C
01390 750* C
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01394 754* C
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01398 758* C
01399 759* C
01400 760* C
01401 761* C
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01460 820* C
01461 821* C
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01489 849* C
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01539 899* C
01540 900* C
01541 901* C
01542 902* C
01543 903* C
01544 904* C
01545 905* C
01546 906* C
01547 907* C
01548 908* C
01549 909* C
01550 910* C
01551 911* C
01552 912* C
01553 913* C
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01559 919* C
01560 920* C
01561 921* C
01562 922* C
01563 923* C
01564 924* C
01565 925* C
01566 926* C
01567 927* C
01568 928* C
01569 929* C
01570 930* C
01571 931* C
01572 932* C
01573 933* C
01574 934* C
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01590 950* C
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01599 959* C
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01630 990* C
01631 991* C
01632 992* C
01633 993* C
01634 994* C
01635 995* C
01636 996* C
01637 997* C
01638 998* C
01639 999* C
01640 1000* C

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***** MAIN *****
01322 647*
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01323 649*
01326 650*
01330 651*
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01332 653*
01333 654*
01334 655*
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01336 657*
01337 658*
01340 659*
01341 660*
01342 661*
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01343 663*
01344 664*
01344 665*
01344 666*
01367 667*
01367 668*
01377 669*
01377 670*
01377 671*
01377 672*
01377 673*
01377 674*
01377 675*
01377 676*
01377 677*
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01377 680*
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01400 695*
01400 696*
01401 697*
01401 698*
01403 699*
01404 700*
01405 701*
01406 702*
01406 703*

C MASS CALCULATIONS
314 DO 315 J=1,MZ
315 Y(J) = 1.0/FW(J)
      TNXL = TNN*XL
      FLMASS = TNXL*AITB*RHO0*DEFINT(Y,DZ,MZ)
      FNMASS = TNXL*HFN*RHOFN*(SROOT*STIP)
      TBMASS = TNXL*PI*STB*(DITB+STB)*RHOTB
      PLMASS = TNXL*SMP*RHO*P*(DITB+2.0*STB*SMP)*PHIM/2.0
      TOTMSS = FLMASS+FNMASS+PLMASS+TBMASS
      FLMASI = FLMASS*32.174
      FNMASI = FNMASS*32.174
      TBMASI = TBMASS*32.174
      PLMASI = PLMASS*32.174
      TOTMSI = TOTMSS*32.174

C
      ATOT = TNXL*(2.0*(HFN+STB)+DITB)
      WRITE(6,320) XL,DITB,STB,TBM,TBMTL,TBMASI,NTBS,
1 HFN,SROOT,STIP,FNM,FMML,FNMASI,NSRAD,
2 FLU,FLUID,FLMASI
      WRITE (6,321) SMP,PLMASI,PRO,PROMTL,
1 TOTMSI,ATOT
320) FORMAT(1H0,15X,17HSYSTEM PARAMETERS,/,16X,17H*****//,
2 6X,35HTUBE LENGTH, XL = ,F10.3,11H FT //,
3 6X,35H INTERNAL DIAMETER, DITB = ,F10.3,11H IN //,
4 6X,35H WALL THICKNESS, STB = ,F10.3,11H IN //,
5 6X,35H MATERIAL, /2A6 //,
6 6X,35H MASS (ALL TUBES), MTB = ,F10.4,11H LBM //,
7 6X,35H NUMBER OF TUBES, NTBS = ,I10 //,
8 6X,35H HEIGHT, HFN = ,F10.3,11H IN //,
9 6X,35H THICKNESS AT ROOT, SROOT = ,F10.3,11H IN //,
1 6X,35H THICKNESS AT TIP, STIP = ,F10.3,11H IN //,
2 6X,35H MATERIAL, /2A6 //,
3 6X,35H MASS (ALL FIN), MFN = ,F10.4,11H LBM //,
4 6X,35H NO. OF FIN SIDES RADIATING = ,I10 //,
5 6X,35H COOLANT FLUID IS, /2A6 //,
6 6X,35H MASS (IN ALL TUBES), MFL = ,F10.4,11H LBM //,
7 6X,35H PROTECTION LAYER THICKNESS, SMP = ,F10.3,11H IN //,
8 6X,35H MASS, MMP = ,F10.3,11H LBM //,
9 6X,35H MATERIAL IS, /2A6 //,
10 6X,35H TOTAL MASS (EXCL. MANIFLD.), MTOT = ,F10.4,11H LBM //,
11 6X,35H TOTAL AREA (SINGLE NORMAL PROJECTION), //,
12 6X,35H ATOT = ,F10.4,11H SQ FT //)

C INITIALIZATION
C
C TUBE
C IF(NRTB.EQ.0) GO TO 341
C
C PFLD = PO*FP(LLC)
C TFLD = TO*FT(LLC)
C RHOFLD = RHOF(PFLD,TFLD)
C XFNU = THCF(RHOFLD,TFLD)/(4.0*THCTB(TINTL))*FNU*REV*PRLNTL
1 *DELTA*DXITB

```

\*\*\*\*\* MAIN \*\*\*\*\*

```

01407 704*      DO 330 J=1,MZ
01412 705*      DO 325 I=2,NRTB1
01415 706*      K = LTB+MZ*(I-1)+J
01416 707*      325 Y(K) = TW(J,I)
01420 708*      LTBJ = LTB+J
01421 709*      330 Y(LTBJ) = (XFNU*FT(J)+4.0*TW(J,2)-TW(J,3))/(XFNU+3.0)
01421 710*      C
01421 711*      C PROTECTION LAYER
01421 712*      C
01421 713*      C
01423 714*      DO 340 J=1,MZ
01426 715*      DO 335 I=2,NRMP1
01431 716*      K = LMP+MZ*(I-2)+J
01432 717*      335 Y(K) = TMP(J,I)
01434 718*      Y(LL4+J) = TMP(J,I)
01435 719*      340 Y(LL5+J) = TMP(J,NRMP)
01435 720*      C
01435 721*      C FIN
01435 722*      C
01435 723*      C
01437 724*      341 DO 350 I=2,NX
01442 725*      DO 350 J=1,MZ
01445 726*      K = MZ*(I-2)+J
01446 727*      350 Y(K) = TEMP(J,I)
01446 728*      C
01446 729*      C FLUID
01446 730*      C
01451 731*      DO 360 J=1,MZ
01454 732*      360 Y(LT+J) = FT(J)
01454 733*      C
01456 734*      IF(NRTB.NE.0) GO TO 364
01460 735*      DO 361 J=1,MZ
01463 736*      K = LL4+J
01464 737*      361 Y(K) = TW(J,I)
01464 738*      C
01466 739*      XX3 = (2.0-PHIM/PI)*(RITB+STB+SMP)
01467 740*      XX4 = 2.0*(SROOT-DXI/2.0*(SROOT-STIP))/(PI*HFN)
01470 741*      XX10 = R10TB*STB*(DITB+STB)/TREF
01471 742*      XX11 = RHOMP*SMP*(DITB+2.0*STB+SMP)*PHIM/(2.0*PI*TREF)
01472 743*      XX12 = RHOFN*DXI*HFN/PI*(SROOT-DXI/4.0*(SROOT-STIP))/TREF
01473 744*      XX13 = PHIM*(RITB+STB+SMP)
01474 745*      XX20 = DXI*HFN/XX13
01475 746*      XX48 = DXI*12.0
01476 747*      FRAD = FRD*XX13/(HFN*PI)
01476 748*      C
01477 749*      364 DO 365 I=1,NEQUS
01502 750*      A(I) = ALIMIT
01503 751*      365 R(I) = RLIMIT
01505 752*      RLIMIT = RLIMIT*5.0
01505 753*      C
01505 754*      C
01506 755*      IF (OR(MRSTRT,2-LTT).GT.8) CALL SHAPEF
01506 756*      C
01506 757*      C
01506 758*      C
01510 759*      TIME = 0.0
01511 760*      DTME = 0.001

```

```

***** MAIN *****
01511 761* C
01512 762*
01513 763* LCT = 1
01514 764* NCTM = 0
01515 765* LIM = 0
01516 766* MODA = 0
01517 767* NCCZ = 0
01518 768*
01519 769* NTRY = 1
01520 770* IFVD = 0
01521 771* IBKP = 1
01522 772* IERR = 0
01523 773* C
01524 774* PRINCIPLE INTEGRATION
01525 775* C
01526 776* CALL RKS(DERIVM,CNTLM,Y,DY,A,R,TIME,DTME,NEQUS,IFVD,IBKP,NTRY,IERR
01527 777* 1 ,DELY,PD,SD,YS,YST,DYST,YSIMP)
01528 778* 9997 RLIMIT = R(1)
01529 779* IF (ISYM.NE.0) GO TO 9999
01530 780* READ (5,GINPT, END = 9999)
01531 781* MRSTRT = AND(MRSTRT,7)
01532 782* GO TO B
01533 783* 9998 FORMAT(IH1)
01534 784* 9999 STOP
01535 785* END

```

END OF COMPILATION: NO DIAGNOSTICS.

OHG:P \*\*\*\*\* MTXINV \*\*\*\*\*

\*\*\*\*\* MTXINV \*\*\*\*\*

QFOR,S ME\*NASAS5.MTXINV,ME\*NASAS5.MTXINV  
FOR S9A-07/13/72-20:57:03 (0,)

SUBROUTINE MTXINV ENTRY POINT 000166

STORAGE USED: CODE(1) 000206; DATA(0) 000037; BLANK COMMON(2) 000000

COMMON BLOCKS:

0003 GRD 003670

EXTERNAL REFERENCES (BLOCK, NAME)

0004 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000016	1056	0001	000076	11L	0001	000031	111G	0001	000052	121G	0001	000055	125G
0001	000117	1366	0001	000122	142G	0001	000137	20L	0000	R	000001	AA	0000	R
0003	003465	CSF	0003	003464	DXX2	0003	003472	DXX21	0003	003463	DZMFN	0003	003410	EBFN
0003	003441	EBMP	0003	003467	EXTIFN	0003	003471	EXTIMP	0003	003466	EXTSFN	0003	003470	EXTSMP
0000	I	000000	I	000011	INJPS	0000	I	000003	I1	0000	I	000006	I2	0000
0000	I	000004	K	0003	003461	LCT	0003	003462	LTT	0003	I	000002	J	0000
0003	R	000000	TRMTX	0003	003446	XX2	0003	003453	ZZ2	0003	003460	MCVRD	0003	003473

```

00101 1* SUBROUTINE MTXINV(N,M)
00101 2* C
00101 3* C THIS SUBROUTINE SOLVES A SYSTEM OF LINEAR ALGEBRIC EQUATIONS BY
00101 4* C MATRIX INVERSION
00101 5* C
00103 6* COMMON /QRD/ TRMTX(30,60),EBFN(5,5),EBMP(5),XX2(5),ZZ2(5),
00103 7* 1 MCVRD,LCT,LTT,DZMFN,
00103 8* 2 DX2,CSF,EXTSFN,EXTIFN,EXTSMP,EXTIMP,DXX21,SS(5,5,5)
00104 9* DO 20 I =1,N
00107 10* AA =TRMTX(I,I)
00110 11* DO 9 J =1,M
00113 12* 9 TRMTX(I,J)=TRMTX(I,J)/AA
00115 13* IF (I.EQ.1) GO TO 11
00117 14* I1 = I-1
00120 15* DO 10 K =1,I1
00123 16* 8 =TRMTX(K,I)
00124 17* DO 10 J =1,M
00127 18* 10 TRMTX(K,J)=TRMTX(K,J) -TRMTX(I,J) * B
00132 19* IF (I.EQ.N) GO TO 20
00134 20* 11 12 = I+1
00135 21* DO 15 K =I2,N
00140 22* 8 =TRMTX(K,I)
00141 23* DO 15 J =1,M
00144 24* 15 TRMTX(K,J)=TRMTX(K,J) -TRMTX(I,J) * B
00147 25* 20 CONTINUE

```

\*\*\*\*\* MTXINV \*\*\*\*\*

00151 26\* RETURN  
00152 27\* END

END OF COMPILATION: NO DIAGNOSTICS.

QH06:P \*\*\*\*\* NUS \*\*\*\*\*

9FOR 5 ME\*NASAS.NUS,ME\*NASAS.NUS  
FOR 59A-07/13/72-20:57:08 (0.)

SUBROUTINE NUS ENTRY POINT 000573

STORAGE USED: CODE(1) 000642; DATA(0) 000114; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 YINT  
0004 NEXP6\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000550	100L	0001	000354	1506	0001	000374	1616	0001	000407	1666	0001	000443	1756	
0001	000243	20L	0001	000456	2026	0001	000513	2116	0001	000526	2166	0001	000324	30L	
0001	000431	40L	0001	000501	50L	0000	R	000015	BETA	0000	R	000021	FORCE		
0000	R	000017	GP	0000	R	000016	GRASH	0000	I	000023	I	0000	000066	INJPS	
0000	R	000004	NUS	0000	R	000010	REY	0000	R	000014	REYNO	0000	R	000022	X

```

00101 1* SUBROUTINE NUS(MACHNO,TATM,CATM,TIN,PRATH,VISATM,RHOATM,
00101 2* STAGX,VERTX,NSRAD,NUS1)
00101 3* C
00101 4* C THIS SUBROUTINE COMPUTES :
00101 5* C A SINGLE NUSSELT NUMBER FOR THE CALCULATION OF THE AERODYNAMIC
00101 6* C HEATING OF THE ORBITER RADIATOR SYSTEM AT THE CURRENT TIME
00101 7* C
00101 8* C REAL MACHNO,NUS1,MACH(4),NUS(4)
00101 9* C DIMENSION REY(4)
00101 10* C REYNO = MACHNO*STAGX*CATM*RHOATM/VISATM
00101 11* C IF (MACHNO .GT. 1.0) GO TO 20
00101 12* C IF (MACHNO .LE. 1.0 .AND. MACHNO .GE. 0.5) GO TO 30
00101 13* C
00101 14* C FREE CONVECTION NUSSELT NUMBER
00101 15* C
00101 16* C BETA = 2.0/(TIN+TATM)
00101 17* C GRASH = 32.174*BETA*(RHOATM/VISATM)**2*VERTX**3.*(TIN-TATM)
00101 18* C GP = GRASH*PRATH
00101 19* C IF (GP .GT. 1.0E-01 .AND. GP .LE. 1.0E+04) FREE = 1.585*(GP)
00101 20* C **0.195
00101 21* C IF (GP .GT. 1.0E+04 .AND. GP .LE. 1.0E+09) FREE = 0.59*(GP)**
00101 22* C 0.25
00101 23* C IF (GP .GT. 1.0E+09) FREE = 0.13*(GP)**0.3333
00101 24* C
00101 25* C LOW SPEED NUSSELT NUMBERS FOR LAMINAR, TURBULENT AND TRANSITION FLOWS
00101 26* C
00101 27* C IF (REYNO .LT. 1.0E+05)
00101 28* C 1 FORCE = 0.332*REYNO**0.5*PRATH**0.3333
00101 29* C IF (REYNO .GT. 1.0E+06)

```

\*\*\*\*\* NUS \*\*\*\*\*

00125 30\* 1 FORCE = 0.0288\*REYN0\*\*0.8\*PRATM\*\*0.33333  
 00127 31\* IF (REYN0 .LE. 1.0E+06 .AND. REYN0 .GE. 1.0E+05)  
 00127 32\* 1 FORCE = 6.78E-05\*PRATM\*\*0.33333\*REYN0\*\*1.238  
 00131 33\* NUS1 = FREE\*STAGX/VERTX\*FORCE  
 00132 34\* GO TO 100  
 00132 35\*

C HIGH SPEED NUSSELT NUMBERS FOR LAMINAR, TURBULENT AND TRANSITION FLOWS  
 C

20 IF (REYN0 .LT. 1.0E+05)  
 1 NUS1 = 0.3751\*PRATM\*REYN0\*\*0.5014  
 IF (REYN0 .GT. 1.0E+06)  
 1 NUS1 = 0.0346\*PRATM\*REYN0\*\*0.7746  
 IF (REYN0 .LE. 1.0E+06 .AND. REYN0 .GE. 1.0E+05)  
 1 NUS1 = 3.339E-04\*PRATM\*REYN0\*\*1.1111  
 GO TO 100

C LAMINAR, TRANSITION AND TURBULENT NUSSELT NUMBERS FOR FLOWS THAT LIE  
 C BETWEEN LOW AND HIGH SPEED MODELS  
 C

30 MACH(1) = 0.4000  
 MACH(2) = 0.6333  
 MACH(3) = 0.8666  
 MACH(4) = 1.1000  
 X = STAGX\*CATM\*RH0ATM/VISATM  
 DO 34 I = 1,4  
 34 REY(I) = MACH(I)\*X  
 IF (REYN0 .GT. 1.0E+06) GO TO 40  
 IF (REYN0 .LT. 1.0E+05) GO TO 50  
 DO 36 I = 3,4  
 36 NUS(I) = 3.339E-04\*PRATM\*REY(I)\*\*1.1111  
 DO 38 I = 1,2  
 38 NUS(I) = 6.78E-05\*PRATM\*\*0.33333\*REY(I)\*\*1.238  
 NUS1 = YINT(MACH,NUS,4,4,MACHNO)

GO TO 100  
 40 DO 44 I = 3,4  
 44 NUS(I) = 0.0346\*PRATM\*REY(I)\*\*0.7746  
 DO 48 I = 1,2  
 48 NUS(I) = 0.0288\*REY(I)\*\*0.8\*PRATM\*\*0.33333  
 NUS1 = YINT(MACH,NUS,4,4,MACHNO)  
 GO TO 100  
 50 DO 54 I = 3,4  
 54 NUS(I) = 0.3751\*PRATM\*REY(I)\*\*0.5014  
 DO 58 I = 1,2  
 58 NUS(I) = 0.322\*REY(I)\*\*0.5\*PRATM\*\*0.33333  
 NUS1 = YINT(MACH,NUS,4,4,MACHNO)  
 100 IF (NSRAD .EQ. 2) NUS1 = 2.0\*NUS1  
 RETURN  
 END

END OF COMPILATION: NO DIAGNOSTICS.

\*\*\*\*\* NUS \*\*\*\*\*



QFOR,S ME\*NASAS,NUSA,ME\*NASAS,NUSA  
FOR S9A-07/13/72-20:57:12 (0,)

FUNCTION NUSA ENTRY POINT 000067

STORAGE USED: CODE(1) 000103; DATA(0) 000024; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NEXP6\$  
0004 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 00024 2L 0001 000043 4L 0001 000057 5L 0000 000016 INJP\$ 0000 R 000000 NUSA  
0000 R 000001 RPD

00101	1*	FUNCTION NUSA(REY,PR,DOL)
00103	2*	REAL NUSA
00104	3*	IF (REY-2300)1,1,2
00107	4*	RPD=REY*PR*DOL
00110	5*	NUSA=3.65+0.0668*RPD/(1+.045*RPD**.6667)
00111	6*	GO TO 5
00112	7*	IF (PR-.1)3,3,4
00115	8*	NUSA=5+.025*(REY*PR)**.8
00116	9*	GO TO 5
00117	10*	NUSA=.023*(REY**.8)*(PR**.3)
00120	11*	RETURN
00121	12*	END

END OF COMPILATION: NO DIAGNOSTICS.

2H06.P \*\*\*\*\* PDERIV \*\*\*\*\*

QFOR.S ME\*NASAS.PDERIV,ME\*NASAS.PDERIV  
 FOR S9A-07/13/72-20:57:13 (0,)

FUNCTION PDERIV ENTRY POINT 000160

STORAGE USED: CODE(1) 000204; DATA(0) 000020; BLANK COMMON(2) 000000

COMMON BLOCKS:

0003 ADBH 000352

EXTERNAL REFERENCES (BLOCK, NAME)

0004 TTIPS  
 0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000036 2L	0001	000114 3L	0003	000001 AL	0003	000161 CONFL	0003	000173 CONFNA
0003	000147 CP	0003	000000 D	0000	R 000001 DM	0003	R 000326 DTL	0003	000244 EFF
0003	000205 EMIS	0003	R 000013 H	0003	R 000231 HAB	0000	000007 INJPS	0003	R 000064 M
0003	000076 NT	0000	R 000000 PDERIV	0003	000052 PIN	0003	000314 GOUT	0003	000217 T8
0003	000026 TH	0003	000040 TIN	0003	R 000270 TL	0003	000340 TOUT	0003	000077 T5TAR
0000	R 000003 T1	0000	R 000002 T2	0003	000135 VSC				

```

00101 1* FUNCTION PDERIV(J,IT,ITST)
00103 2* PARAMETER NTP = 10
00104 3* PARAMETER NTP1 = NTP+1
00105 4* REAL M
00106 5* COMMON /ADBH/ D,AL(NTP),H(NTP1),TH(NTP),TIN(NTP),PIN(NTP),
00106 6* M(NTP),NT,TSTAR(NTP,3),VSC(NTP),CP(NTP),CONFL(NTP),
00106 7* CONFNA(NTP),EIS(NTP),TB(NTP),HAB(NTP1),EFF(NTP,2),TL(NTP,2),
00106 8* GOUT(NTP),DTL(NTP),TOUT(NTP)
00107 9* IF(J) ,2,3
00112 10* CALL TTIPS(IT,ITST,1.01*HAB(IT),2*H(IT+1)-HAB(IT+1),DM,T2)
00113 11* PDERIV=(T2-TL(IT+1,1))-DTL(IT)/(,01*HAB(IT))
00114 12* RETURN
00115 13* CALL TTIPS(IT,ITST,HAB(IT),2.0*H(IT+1)-1.01*HAB(IT+1),DM,T2)
00116 14* CALL TTIPS(IT+1,ITST,1.01*HAB(IT+1),2*H(IT+2)-HAB(IT+2),T1,DM)
00117 15* PDERIV=(T2-T1-DTL(IT))/(,01*HAB(IT+1))
00120 16* RETURN
00121 17* CALL TTIPS(IT+1,ITST,HAB(IT+1),2*H(IT+2)-1.01*HAB(IT+2),T1,DM)
00122 18* PDERIV=(T1-T2-T1-DTL(IT))/(,01*HAB(IT+2))
00123 19* RETURN
00124 20* END
  
```

END OF COMPILATION: NO DIAGNOSTICS.

\*\*\*\*\* POERIV \*\*\*\*\*  
OHOG.P \*\*\*\*\* PF/CFHE \*\*\*\*\*

\*\*\*\*\* PF/CFHE \*\*\*\*\*

QFOR'S MERNASAS,PF/CFHE,ME\*NASAS,PF/CFHE  
FOR SPA-07/13/72-20:57:16 (0\*)

FUNCTION PF ENTRY POINT 000047

STORAGE USED: CODE(1) 000052; DATA(0) 000026; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000005 A 0000 R 000012 ALPHA 0000 R 000002 A1 0000 R 000003 B1 0000 R 000004 C  
 0000 000017 INJPS 0000 R 000000 PF 0000 R 000001 R 0000 R 000006 RHOX 0000 R 000007 TX  
 0000 R 000010 V 0000 R 000011 VM2

00101 1\* FUNCTION PF(RHO,T)

00101 2\* C  
 00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES:  
 00101 4\* C PRESSURE AS A FUNCTION OF DENSITY (SLUG/CU.FT) AND  
 00101 5\* C TEMPERATURE (R) OF HELIUM  
 00101 6\* C UNITS LBF/SQ.FT  
 00101 7\* C  
 00101 8\* DATA R=A1,B1,C,A /2077.02,136.9595,3.5002295E-03,10.000658,1.49610  
 00103 9\* 13E-02/  
 00111 10\* RHOX = RHO\*515.4275  
 00112 11\* TX = T/1.8  
 00113 12\* V = 1.0/RHOX  
 00114 13\* VM2 = RHOX\*RHOX  
 00115 14\* ALPHA = C\*RHOX/(TX\*\*3)  
 00116 15\* PF = (R\*TX\*(1.0-ALPHA)\*(V\*B1)-A1\*(1.0-A\*RHOX))\*VM2\*.08846E-02  
 00117 16\* RETURN  
 00120 17\* END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG:P \*\*\*\*\* PF/CFNAK \*\*\*\*\*

\*\*\*\*\* PF/CFNAK \*\*\*\*\*

QFOR,S ME\*NASAS,PF/CFNAK,ME\*NASAS,PF/CFNAK  
FOR S9A-07/13/72-20:57:18 (0,)

FUNCTION PF ENTRY POINT 000030

STORAGE USED: CODE(1) 000040; DATA(0) 000015; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 CAPPA  
0004 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000004 C 0003 R 000000 CAPPA 0000 000010 INJPS 0000 R 000003 P0  
0000 R 000005 TK 0000 R 000001 X1 0000 R 000002 X2

00101 1\* FUNCTION PF(RHO,T)

00101 2\* C  
00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
00101 4\* C PRESSURE AS A FUNCTION OF DENSITY (SLUG/CU.FT) AND  
00101 5\* C TEMPERATURE (R) OF NAK 78.6  
00101 6\* C UNITS LBF/SQ.FT  
00101 7\* C  
00101 8\* C DATA X1,X2 /58.773064\*-0.008433/P0 /2116.224/  
00101 9\* C C = (X1\*X2\*T)/32.174  
00110 10\* C TK = CAPPA(RHO,T)  
00111 11\* C PF = (RHO/C-1.0)/TK\*P0  
00112 12\* C RETURN  
00113 13\* C END

END OF COMPILATION: NO DIAGNOSTICS.

PHOG,P \*\*\*\*\* PF/CFSIL \*\*\*\*\*

\*\*\*\*\* PF/CFSIL \*\*\*\*\*

QFOR'S ME\*NASAS5.PF/CFSIL,ME\*NASAS5.PF/CFSIL  
FOR S9A-07/13/72-20:57:20 (0.)

FUNCTION PF ENTRY POINT 000155

STORAGE USED: CODE(1) 000162; DATA(0) 000072; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 ALOG  
0004 SORT  
0005 NWDJ5  
0006 NIO25  
0007 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000	000030	IF	0000 R 000022 A	0000 R 000001 A1	0000 R 000002 A2	0000 R 000003 A3
0000	R 000004	A4	0000 R 000005 A5	0000 R 000023 B	0000 R 000006 B1	0000 R 000007 B2
0000	R 000010	B3	0000 R 000011 B4	0000 R 000012 B5	0000 R 000024 C	0000 R 000013 C1
0000	R 000014	C2	0000 R 000017 D1	0000 000061 INJPS	0000 R 000027 P	0000 R 000000 PF
0000	R 000025	RH01	0000 R 000020 THETA	0000 R 000021 THETA1	0000 R 000015 T0	0000 R 000016 T01
0000	R 000026	Z				

FUNCTION PF(RHO,T)

```

1* 00101
2* 00101
3* 00101
4* 00101
5* 00101
6* 00101
7* 00101
8* 00101
9* 00103
10* 00103
11* 00103
12* 00123
13* 00124
14* 00125
15* 00126
16* 00127
17* 00130
18* 00131
19* 00132
20* 00133
21* 00134
22* 00134
23* 00141
24* 00141
25* 00142
26* 00143

```

THIS FUNCTION SUBPROGRAM COMPUTES :  
PRESSURE AS A FUNCTION OF DENSITY (SLUG/CU.FT) AND TEMPERATURE  
(R) OF DOW CORNING 200 SILICON OIL (1 CS)  
UNITS LBF/SQ.FT  
TEMPERATURE ,GE, 359.67 AND .LE, 859.67  
TEMP. GE, 360.67 AND LE, 860.67  
DATA A1,A2,A3,A4,A5 /12.35,2.98333,1.1,-0.48333,0.1/ ,B1,B2,B3,B4,  
185 /-1.5,-0.01333,-1.18,0.57333,-0.1/ ,C1,C2 /0.7767,-0.0288/ ,  
2T0,T01,DT /559.67,609.67,50.0/  
THETA = (T-T0)/DT  
THETA1 = (T-T01)/DT  
A = (((A5\*THETA+A4)\*THETA+A3)\*THETA+A2)\*THETA+A1)  
B = (((B5\*THETA+B4)\*THETA+B3)\*THETA+B2)\*THETA+B1)  
C = C1+C2\*THETA1  
RH01 = RH032\*.174/62.42  
Z = A\*A+2.0\*B\*ALOG(RH01/C)\*1000.0  
P = (-A\*SQRT(Z))\*1000.0/B  
PF = (P+14.696)\*144.0  
IF (T.LT.360.67.OR.T.GT.860.67.OR.PF.GT.146116.224.OR.PF.GT.110116  
1.224.AND.T.LT.460.67) WRITE(6,1) T,PF  
1 FORMAT (JH0,3HPRESSURE OF SILICON OIL, OUT OF RANGE, T = ,F10.5,  
16H, P = ,F15.5//)  
RETURN  
END

\*\*\*\*\* PF/CFSIL \*\*\*\*\*

END OF COMPILATION: NO DIAGNOSTICS.

QHDG:P \*\*\*\*\* POLY \*\*\*\*\*

DATE 071372

PAGE 2

\*\*\*\*\* POLY \*\*\*\*\*

QFOR,S ME\*NASAS.POLY,ME\*NASAS.POLY  
FOR S9A-07/13/72-20:57:23 (0,)

FUNCTION POLY ENTRY POINT 000036

STORAGE USED: CODE(1) 000044; DATA(0) 000015; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000012 1076 0000 000003 INJPS 0000 I 000001 L 0000 R 000000 POLY

```

00101 1*          FUNCTION POLY(N,A,X)
00101 2*          C
00101 3*          C THIS SUBROUTINE EVALUATES POLYNOMIAL
00101 4*          C POLY = A(1)+A(2)*X+A(3)*X**2+.....+A(N)*X**(N-1)
00101 5*          C
00103 6*          DIMENSION A(N).
00104 7*          POLY = 0.
00105 8*          L = N
00106 9*          DO 1 K=1,N
00111 10*         POLY = POLY*X+A(L)
00112 11*         1 L = L-1
00114 12*         RETURN
00115 13*         END

```

END OF COMPILATION: NO DIAGNOSTICS.

GH06:P \*\*\*\*\* @INCID \*\*\*\*\*



\*\*\*\*\* GINCID \*\*\*\*\*

QFOR S ME\*NASAS.GINCID,ME\*NASAS.GINCID  
FOR S9A-07/13/72-20:57:27 (0,)

SUBROUTINE GINCID ENTRY POINT 001001

STORAGE USED: CODE(1) 001027; DATA(0) 000665; BLANK COMMON(2) 000000

COMMON BLOCKS:

0003 GIN 001610  
0004 TC 000042

EXTERNAL REFERENCES (BLOCK, NAME)

0005 YINT  
0006 SHADE  
0007 DEFINT  
0010 NRBU\$  
0011 NI02\$  
0012 N\$P\$  
0013 NI01\$  
0014 NRE\*\$  
0015 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000036	10L	0001	000036	123G	0001	000064	142G	0001	000124	160G	0001	000132	165G				
0001	000151	173G	0001	000057	20L	0001	000166	203G	0001	000073	22L	0001	000273	225G				
0001	000337	2403	0001	000354	247G	0001	000404	263G	0001	000230	30L	0001	000547	301G				
0001	000255	32L	0001	000730	324G	0001	000741	330G	0001	000313	36L	0001	000325	38L				
0001	000507	58L	0001	000633	70L	0001	000643	80L	0004	000003	AEMF	0000	000000	ANG				
0007	R	000000	DEFINT	0000	I	000505	I	000001	ICASE	0000	I	000603	IFILE	ANG				
0000	I	000604	IP	0004	I	000000	ITAPE	0000	I	000610	K	0000	I	000614	KO			
0000	I	000615	K1	0000	I	000612	M1	0000	I	000613	M2	0000	I	000606	NS			
0003	I	001605	NSRC	0003	I	001604	NTM	0000	I	000607	N1	0000	R	000004	PHIN			
0000	R	000343	QA	0000	R	000077	GALB	0003	R	000454	GIFN	0000	R	000463	GP			
0000	R	000151	QPLN	0000	R	000223	GS	0003	R	000144	GSFN	0000	R	000025	GSUN			
0003	001606	QTO	0003	R	000000	TM	0004	R	000002	TO	0003	R	001607	TX	0005	R	000000	YINT

SUBROUTINE GINCID(MSTOTR,PHN)

00101 1\* C  
00101 2\* C  
00101 3\* C  
00101 4\* C  
00101 5\* C  
00101 6\* C  
00101 7\* C  
00101 8\* C  
00101 9\* C  
00101 10\* C  
00101 11\* C

THIS SUBROUTINE PROVIDES :  
1- DATA TRANSFER FROM THE SRI INCIDENT RADIATION COMPUTER PROGRAM  
2- AVERAGING OF RADIANT FLUX OVER CIRCUMFERENCE OF TUBE CROSS-SECTION  
PARAMETER NTP = 10  
PARAMETER MAXSDI = 20  
PARAMETER MAXSDI = MAXSDI\*1  
COMMON /GIN/ TM(100),GSFN(100,2),QIFN(100,2),QSTB(100,2),

\*\*\*\*\* QINCID \*\*\*\*\*

```

00106 12* 1 QITB(100,2),NTM,NSRD,QT0,TX
00107 13* COMMON /TC/ ITAPE,ICASE,TO,ABEMF,PHIN(NTP,3)
00110 14* DIMENSION ANG(MAXSD1),OSUN(MAXSD1,2),GALB(MAXSD1,2),
00110 15* OPLN(MAXSD1,2),QS(4,MAXSID),QA(4,MAXSID),QP(4,MAXSID)
00111 16* IFILE = ITAPE
00112 17* IF (ITAPE.LE.0) IFILE = NSRD+2
00114 18* IP = 1
00115 19* IF (ABS(PHN).GT.0.0) IP = 2
00117 20* IF (ICASE.LE.1) GO TO 20
00121 21* TM(1) = -(10.0**30)
00122 22* DO 16 I = 2,ICASE
00125 23* TM(2) = TM(1)
00126 24* READ (IFILE) NS,TM(1)
00132 25* IF (TM(1).GT.TM(2)) GO TO 10
00134 26* CONTINUE
00136 27* BACKSPACE IFILE
00137 28* IF (NSTOTR.EQ.2) GO TO 30
00141 29* DO 21 I = 1,2
00144 30* READ (IFILE)
00147 31* N1 = 2
00150 32* READ (IFILE) NS,TM(1)
00154 33* N1 = N1+1
00155 34* IF (TM(1).LT.TO.AND.N1.LT.NTM-1) GO TO 22
00157 35* DO 24 I = 1,3
00164 36* BACKSPACE IFILE
00167 38* DO 26 I = 1,4
00202 39* READ (IFILE) NS,TM(1),QS(I,K),QA(I,K),QP(I,K),K=1,NS)
00205 40* OSUN(K,1) = YINT(TM,QS(I,K),4,4,TO)
00206 41* GALB(K,1) = YINT(TM,QA(I,K),4,4,TO)
00207 42* OPLN(K,1) = YINT(TM,QP(I,K),4,4,TO)
00211 43* TM(1) = 0.0
00212 44* NTM = 1
00213 45* IF (ICASE.GT.1.OR.NSTOTR.EQ.1) GO TO 32
00215 46* READ (IFILE) NS
00220 47* BACKSPACE IFILE
00221 48* NP = NS/2
00222 49* M1 = (NP-1)/2
00223 50* M2 = NP/2
00224 51* DO 34 I = 0,NS
00227 52* ANG(I+1) = FLOAT(I*360+180)/NS
00231 53* IF (IP.EQ.1) GO TO 38
00233 54* K0 = 0
00234 55* K0 = K0+1
00235 56* IF (ANG(K0)+ANG(1).LE.PHN) GO TO 36
00237 57* DO 90 I = 1,NTM
00242 58* IF (NSTOTR.EQ.2) READ (IFILE) NS,TM(I),OSUN(K,1),GALB(K,1),
00242 59* OPLN(K,1),K=1,NS)
00255 60* IF (IP.EQ.1) GO TO 58
00257 61* OSUN(NS+1,1) = OSUN(1,1)
00260 62* GALB(NS+1,1) = GALB(1,1)
00261 63* OPLN(NS+1,1) = OPLN(1,1)
00262 64* DO 54 K = 1,NS
00265 65* K1 = MIN(MAX(MOD(K+K0-2,NS)+1),NS-2)
00266 66* PANG = ANG(K)+PHN
00267 67* IF (PANG.GE.ANG(NS+1)) PANG = PANG-360.0
00271 68* OSUN(K,2) = YINT(ANG(K1),OSUN(K1,1),4,4,PANG)

```

\*\*\*\*\* QINCID \*\*\*\*\*

```

00272 69*
00273 70*
00275 71*
00276 72*
00277 73*
00300 74*
00303 75*
00304 76*
00306 77*
00307 78*
00307 79*
00310 80*
00311 81*
00312 82*
00313 83*
00314 84*
00314 85*
00317 86*
00320 87*
00322 88*
00323 89*
00326 90*
00327 91*
00332 92*
00333 93*
00334 94*
00335 95*
00340 96*
00341 97*

54
58

70
80
90

100

***** QINCID *****
QALB(K,2) = YINT(ANG(K1),QALB(K1,1),4,4,PANG)
QPLN(K,2) = YINT(ANG(K1),QPLN(K1,1),4,4,PANG)
CALL SHADE(QSUN(1,IP),NS)
CALL SHADE(QALB(1,IP),NS)
CALL SHADE(QPLN(1,IP),NS)
DO 90 J = 1,NSRD
N1 = NP*(J-1)+1
IF (M1.EQ.M2) GO TO 70
QSFN(I,J) = YINT(ANG(M1),QSUN(N1+M1-1,IP),4,4,90.0)
QIFN(I,J) = YINT(ANG(M1),QALB(N1+M1-1,IP),4,4,90.0)+
YINT(ANG(M1),QPLN(N1+M1-1,IP),4,4,90.0)
GO TO 80
QSFN(I,J) = QSUN(N1+M1,IP)
QIFN(I,J) = QALB(N1+M1,IP)+QPLN(N1+M1,IP)
QSTB(I,J) = DEFINT(QSUN(N1,IP),0.5,NP)
QITB(I,J) = DEFINT(QALB(N1,IP),0.5,NP)+
DEFINT(QPLN(N1,IP),0.5,NP)
REWIND IFILE
IF (MSTOTR.EQ.2) RETURN
NTM = 4
DO 100 I = 2,4
TM(I) = 20*(I-1)
DO 100 J = 1,NSRD
QSFN(I,J) = QSFN(1,J)
QIFN(I,J) = QIFN(1,J)
QSTB(I,J) = QSTB(1,J)
QITB(I,J) = QITB(1,J)
RETURN
END

```

END OF COMPILATION: NO DIAGNOSTICS.

QHDG.P \*\*\*\*\* GRAD \*\*\*\*\*

\*\*\*\*\* GRAD \*\*\*\*\*

QFOR,S ME\*NASAS5,GRAD,ME\*NASAS5,GRAD  
 FOR S9A-07/13/72-20:57:51 (0,)

SUBROUTINE GRAD ENTRY POINT 000772

STORAGE USED: CODE(1) 001013; DATA(0) 001077; BLANK COMMON(2) 001115

COMMON BLOCKS:

0003 GRD 003721  
 0004 GIN 001610  
 0005 ABSRST 000601  
 0006 SSF 000010  
 0007 AVGABS 000251

EXTERNAL REFERENCES (BLOCK, NAME)

0010 INTERP  
 0011 YINT  
 0012 EMIT  
 0013 AVGEVT  
 0014 ABSORB  
 0015 TRMATX  
 0016 MIXINV  
 0017 EXITAV  
 0020 DEFNT  
 0021 NERR2\$  
 0022 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000245	100L	0001	000010	1226	0001	000012	1266	0001	000046	1356	0001	000055	1426							
0001	000073	147G	0001	000123	156G	0001	000112	2L	0001	000304	206G	0001	000306	211G							
0001	000346	225G	0001	000362	235G	0001	000366	241G	0001	000422	252G	0001	000434	256G							
0001	000443	261G	0001	000524	273G	0001	000536	277G	0001	000632	313G	0001	000633	316G							
0001	000645	324G	0001	000702	337G	0001	000733	346G	0001	000735	352G	0001	000332	500L							
0001	000344	550L	0001	000360	600L	0001	000407	700L	0001	000641	810L	0001	000673	820L							
0001	000717	910L	0006	000002	ALPFN	0005	000000	ALPHFN	0005	000226	ALPHMP	0002	000562	AUX2							
0000	R	000163	AX1	0000	R	000327	AX2	0000	R	000637	AX4	0002	000752	COB							
0002	R	001107	COH	0002	000574	CONFN	0002	000740	CONMP	0002	001070	CP0	0003	R	003465	CSF					
0020	R	000000	DEFNT	0002	001072	DELTA	0002	000764	DOB	0002	001110	DXI	0002	001053	DXIMP						
0002	R	001052	DXITB	0003	R	003464	DXX2	0003	003472	DXX21	0002	001051	DZ	0003	003463	DZMFN					
0003	R	003410	EBFN	0003	R	003441	EBMP	0012	R	000000	EMIT	0005	000461	EMITFN	0005	000512	EMITMP				
0003	R	003467	EXTIFN	0003	R	003471	EXTIMP	0003	R	003466	EXTSFN	0003	R	003470	EXTSMP	0005	R	000517	EXTVCT		
0002	R	001103	FENTH	0002	001045	FF0	0002	001101	FLUX	0002	001041	FMACH	0002	001040	FNU	0002	001040	FNU			
0002	001046	FOF	0002	000012	FP	0002	001042	FPR	0002	001037	FR	0002	001050	FRAD	0002	001044	FRM				
0002	001111	FRD	0002	001036	FRE	0002	001104	FREJ	0002	001043	FRL	0002	001044	FRM	0002	001047	FZ				
0002	000036	FT	0002	000024	FW	0002	001113	FXHW	0002	001112	FXOH	0000	I	001024	IJK	0000	I	001016	K		
0002	001105	HFN	0000	I	001006	I	0002	001003	IFLOW	0000	I	001007	III	0000	I	001024	IJK	0000	I	001016	K
0000	I	001022	IL	0000	I	001021	KL	0000	I	001017	L	0002	001024	LC	0003	I	003461	LCT			
0000	I	001025	KJL	0000	I	001021	KL	0000	I	001017	L	0002	001012	LL2	0002	001013	LL3				
0002	001032	LIM	0002	001031	LIMWRT	0002	001011	LL1	0002	001016	LL6	0002	001017	LL7	0002	001006	LMP				
0002	001014	LL4	0002	001015	LL5	0002	001016	LL6	0002	001016	LL6	0002	001017	LL7	0002	001006	LMP				

```

0002 001004 LT
0003 I 003460 MCVRD
0002 001034 NCCZ
0002 001027 NM1
0002 001021 NRTB1
0002 001062 PH1M
0004 R 001274 GIT3
0004 R 001144 QSFH
0002 001067 RH00
0003 R 003873 SS
0007 R 000100 TFIH
0000 R 000000 TAPP
0002 000050 TW
0002 001066 W0
0002 000000 XIIB
0000 R 001014 XACFN
0007 000031 XXXMP
0002 001055 XX3
0000 R 001010 ZFN
0002 001005 LTB
0002 001030 MM1
0002 001035 NCONV
0002 I 001001 NRMP
0004 I 001605 NSRD
0002 001071 PI
0000 R 000012 GMP
0004 R 000764 QSTB
0002 001073 RLIMIT
0003 R 003670 S5TT
0002 001077 TI
0007 R 000244 TPR
0004 R 001607 TX
0000 R 001012 XCFN
0002 001106 XL
0000 R 001014 XACFN
0007 000031 XXXMP
0002 001057 XX10
0002 001056 XX4
0000 R 001011 ZMP
0002 001007 LTBWZ
0002 001033 MOD
0002 001022 NCTL
0002 001020 NRMP1
0004 I 001604 NTM
0002 001065 PO
0002 001102 QREF
0004 R 001606 QTO
0002 001100 RTEND
0006 000001 T
0002 001076 TINTL
0002 001075 TREF
0000 R 001064 TO
0000 R 001013 XCMP
0002 001054 XRE
0007 R 000062 XXFN
0002 001060 XX11
0011 R 000000 YINT
0003 R 003453 ZZZ
0002 001010 LTB2MZ
0002 001025 MSTOTR
0002 001026 NCTM
0002 001022 NRMP2
0002 I 000776 NX
0000 R 000017 GFN
0002 R 000360 GRFN
0002 001074 RDTWRT
0006 000000 R2
0002 001114 RHOFN
0006 000000 R2
0002 R 000132 TEMP
0002 R 000276 TMP
0000 R 001003 TSOL
0006 000007 WWB
0002 000005 XIIMP
0006 R 000004 XSPM
0007 000071 XXXFN
0003 R 003446 XX2
0002 R 000550 ZETA

```

SUBROUTINE GRAD

```

C
C THIS SUBROUTINE COMPUTES :
C 1- INVERSION OF TRANSFER MATRIX
C 2- SOLUTION OF RADIOISOTOPY EQUATION
C 3- GRID MAPPING

```

```

NET RADIANT HEAT LOSS, AVERAGED OVER REPRESENTATIVE AREA ELEMENT
(OVER BOTH SIDES OF THE FIN, BOTH HALVES OF THE TUBE)

```

```

COMMON XIIB(5),XIWP(5),FP(10),FW(10),FT(10),TW(10,5),TEMP(10,10),
  TMP(10,5),ORFN(10,10),ORMP(10),XIFN(10),ZETA(10),AUX2(10),
  CONFN(10,10),CONMP(10),COB(10),DOB(10)
COMMON NX,MZ,NRTB,NRMP,NCTL,IFLOW,LT,LTB,LMP,LIBMZ,LTB2MZ,LL1,
  LL2,LL3,LL4,LL5,LL6,LL7,NRXP1,NRTB1,NRMP2,NEQUS,LC,MSTOTR,
  NCTM,NM1,NM2,LIM,MOD,NCCZ,NCONV
COMMON FRE,FR,FNU,FMACH,FR,FRL,FRM,FFO,FOF,FZ,FRAD,DZ,DXITB,DXIMP,
  XRE,XX3,XX4,XX10,XX11,XX12,PHIM,PHIF,TO,PO,W0,RH00,CPO,PI
  DELTA,RLIMIT,RDTWRT,TREF,TINTL,TI,RTEND,FLUX,QREF,FENTH,
  FREJ,HFN,XL,COH,DXI,FR,FXOH,FXHM,RHOFN
COMMON /GRD/ TRMTX(30,60),EBFN(5,5),EBMP(5),XX2(5),ZZ2(5),
  MCVRD,LCT,LT,DZM,N
  DXX2,CSF,EXTSFN,EXTIFN,EXTSMP,EXTIMP,DXX21,SS(5,5),
  S5TT(5,5)
COMMON /GIN/TM(100), QSFN(100,2), QIFN(100,2), QSTB(100,2),
  GITB(100,2), NTM, NSRD, QTO, TX
COMMON /ABSRST/ALPHFN(5,5,6),ALPHMP(5,31),EMITFN(5,5),EMITMP(5),
  EXTVC(150)
COMMON /SSF/R2, T, ALPFN,R1,XSPM,XSMP,WWA,WWB
COMMON/AVGABS/XXMP(5,5),XXX4P(5,5),XXFN(7),XXXFN(7),
  TFN(10,10),TPR(5)
DIMENSION TAPP(10),GMP(15),GFN(10,10),
  AX1(100),AX2(100),AX3(100),AX4(100)

```

```

***** GRAD *****
00114 35*
00114 36*
00117 37*
00121 38*
00124 39*
00125 40*
00130 41*
00130 42*
00133 43*
00133 44*
00134 45*
00137 46*
00141 47*
00144 48*
00145 49*
00146 50*
00151 51*
00151 52*
00151 53*
00154 54*
00155 55*
00160 56*
00161 57*
00162 58*
00163 59*
00165 60*
00166 61*
00167 62*
00170 63*
00171 64*
00173 65*
00174 66*
00174 67*
00175 68*
00175 69*
00176 70*
00177 71*
00200 72*
00201 73*
00202 74*
00203 75*
00205 76*
00210 77*
00213 78*
00215 79*
00216 80*
00220 81*
00221 82*
00221 83*
00222 84*
00223 85*
00223 86*
00224 87*
00227 88*
00231 89*
00232 90*
00233 91*

***** GRAD *****
DATA TSOL,TALB/10*00.0*480.0/
C IF(LTT,GT,1) GO TO 910
DO 1 J = 1,5
GMP(J) = 0.0
DO 1 I = 1,5
1 GFN(J,I) = 0.0
C
C CALL INTERP(NX,MZ,XIFN,ZETA,TEMP,5,5,XX2,ZZ2,TFIN)
00 5 J = 1,MZ
5 TMRP(J) = TMP(J,MRMP)
DO 10 J = 1,5
TPR(J) = YINT(ZETA,TMRP,MZ,3,ZZ2(J))
EBMP(J) = TPR (J)**4
DO 10 I = 1,5
10 EBFN(J,I) = TFIN(J,I)**4
C
III = 1
2 DO 3 J = 1,NTM
AX1(J) = QIFN(J,III)
AX2(J) = QSFN(J,III)
AX3(J) = QITB(J,III)
3 AX4(J) = QSTB(J,III)
EXTFN = YINT(TM,AX1,NTM,3,IX)/QTO
EXTSFN = YINT(TM,AX2,NTM,3,IX)/QTO
EXTIMP = YINT(TM,AX3,NTM,3,IX)/QTO
EXTSMP = YINT(TM,AX4,NTM,3,IX)/QTO
ZFN = EXTSFN+EXTIFN
ZMP = EXTSMP+EXTIMP
XCFCN = (EXTSFN*EMIT(TSOL)+EXTIFN*EMIT(TALB))/ZFN
XCFCMP = (EXTSMP*EMIT(TSOL)+EXTIMP*EMIT(TALB))/ZMP
C
60 TO (100,500),LCT
C
100 CALL AVGEMT(T0)
XXCFN = (EXTSFN*XXFN(7)+EXTIFN*XXFN(6))/ZFN
XXCFCMP = (EXTSMP*XXFN(7)+EXTIMP*XXFN(6))/ZMP
CALL ABSORB(T0,XCFN,XCFCMP,XXCFN,XXCFCMP)
CALL TRMATX
IF (LTT,NE,0) GO TO 500
DO 115 I = 1,30
DO 110 J = 31,60
110 TRMTX(I,J) = 0.0
K
115 TRMTX(I,K) = 1.0
CALL MTXINV(30,60)
LCT = 2
C
500 CALL EXITAV
60 TO (550,600),LCT
C
550 DO 555 I = 1,30
555 TRMTX(I,31) = EXTVCT(I)
CALL MTXINV (30,31)
L = 31
60 TO 700

```

```

***** GRAD *****
00233 92*
00234 93*
00237 94*
00240 95*
00243 96*
00244 97*
00246 98*
00250 99*
00250 100*
00250 101*
00251 102*
00254 103*
00255 104*
00260 105*
00263 106*
00264 107*
00266 108*
00267 109*
00271 110*
00271 111*
00272 112*
00275 113*
00276 114*
00301 115*
00302 116*
00304 117*
00304 118*
00304 119*
00307 120*
00310 121*
00310 122*
00312 123*
00315 124*
00320 125*
00320 126*
00323 127*
00326 128*
00330 129*
00332 130*
00334 131*
00335 132*
00335 133*
00336 134*
00341 135*
00342 136*
00344 137*
00345 138*
00350 139*
00351 140*
00354 141*
00357 142*
00360 143*

C 600 DO 610 I = 1,30
Z = 0.0
DO 605 J = 1,30
K = J+30
605 Z = Z+TRMTX(I,K)*EXTVCT(J)
610 TRMTX(I,1) = Z
L = 1

C 700 DO 730 J=1,5
KL = J+25
DO 720 IL=1,5
DO 715 JL=1,5
K = (IL-1)*5+JL
715 AX1(JL) = CSF*SS(IL,JL)*TRMTX(K,L)
AX3(IL) = SST(J,IL)*TRMTX(IL+25,L)
720 AX2(IL) = DEFNT(AX1,0.25,5)
1 GMP(IJ) = -(ZMP*DEFNT(AX2,DX2,5)+AX2(1)*DX2/2.0+DEFNT(AX3,
DO 730 I=1,5
IJK = (I-1)*5+J
DO 725 JL=1,5
KJL = 25+JL
725 AX1(JL) = SST(I,J,JL)*TRMTX(KJL,L)
730 GFN(I,J,1) = -(ZFN*DEFNT(AX1,0.25,5)-TRMTX(IJK,L))*GFN(J,I)

C CALL INTERP(5,5,XX2,ZZ2,GFN,NX,MZ,XIFN,ZETA,GRFN)
IF(MCVRO,LE.0) GO TO 810
DO 800 I = 1,MCVRO
DO 800 J = 1,MZ
800 GRFN(J,I) = 0.0

C 810 DO 815 J = 1,MZ
815 GRMP(J) = YINT(ZZ2,OMP,5,3,ZETA(J))
IF (III,EG.2) GO TO 820
IF (NSRD,EG.1) GO TO 820
III = 2
GO TO 2

C 820 DO 825 J = 1,MZ
K = MCVRO+1
825 GRFN(J,K) = GRFN(J,K)*XSPM
RETURN
910 DO 915 J=1,MZ
GR4P(J) = 0.0
DO 915 I=1,NX
915 GRFN(J,I) = 0.0
RETURN
END

```

END OF COMPILATION: NO DIAGNOSTICS.

\*\*\*\*\* GRAD \*\*\*\*\*  
\*\*\*\*\* QHDS/P \*\*\*\*\* REFP \*\*\*\*\*



SUBROUTINE REFP ENTRY POINT 000316

STORAGE USED: CODE(1) 000367; DATA(0) 000163; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

- 0003 CPAIR
- 0004 POLY
- 0005 NEXP65
- 0006 EXP
- 0007 NERRJ5

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	R	000076	2L	0001	000214	20L	0001	000065	223G	0001	000154	3L	0001	000172	4L
0000	R	000066	A0	0000	R	000067	A1	0000	R	000100	A10	0000	R	000070	A2
0000	R	000071	A3	0000	R	000072	A4	0000	R	000073	A5	0000	R	000075	A7
0000	R	000076	A8	0000	R	000077	A9	0000	R	000064	CF1	0000	R	000000	CPAIR
0000	R	000056	CP1	0000	R	000057	CP2	0000	R	000060	CP3	0000	R	000110	DH
0000	R	000063	DT	0000	R	000104	ELEVM	0000	R	000106	GEOH	0000	R	000007	GH
0000	R	000031	GT	0000	R	000052	G0	0000	R	000000	H	0000	I	000144	INJPS
0000	I	000111	J	0000	R	000102	MWT	0000	R	000041	P	0000	R	000000	POLY
0000	R	000116	PSL	0000	R	000115	PZ	0000	R	000054	RU	0000	R	000020	T
0000	R	000112	TMB	0000	R	000103	TMK	0000	R	000062	T0	0000	R	000114	Z

00101	1*	SUBROUTINE REFP (ELEV,REFTP,REFPR,REFVIS,REFRHO,REFK,REFCP,
00101	2*	REFGAM)
00101	3*	1
00101	4*	C THIS SUBROUTINE COMPUTES :
00101	5*	C PROPERTIES OF ATMOSPHERIC AIR EVALUATED AT THE HIGH SPEED
00101	6*	C REFERENCE TEMPERATURE
00101	7*	C ELEVATION IN FEET
00101	8*	C REFERENCE TEMP. IN DEGREES R.
00101	9*	C PRANDTL NUMBER DIMENSIONLESS
00101	10*	C DYN. VISCOSITY IN LBM/ FT. SEC.
00101	11*	C DENSITY IN LBM/CUBIC FOOT
00101	12*	C THERMAL COND. IN BTU/SEC FT DEG R
00101	13*	C SPECIFIC HEAT IN BTU/LBM DEG R
00101	14*	C DIMENSION H(7),GH(9),T(9),GT(8),P(9)
00103	15*	C DATA H(1),H(2),H(3),H(4),H(5),H(6),H(7) /0.0,1.0,-1.5731262E-07,2.
00104	16*	C 14656553E-14,-3.8667054E-21,6.0621354E-28,-9.5013649E-35/,GH(1),GH(
00104	17*	C 22),GH(3),GH(4),GH(5),GH(6),GH(7),GH(8),GH(9) /0.0,11.0,20.0,32.0,4
00104	18*	C 37.0,52.0,61.0,79.0,90.0/T(1),T(2),T(3),T(4),T(5),T(6),T(7),T(8),T
00104	19*	C 4(9) /288.15,216.65,216.65,228.65,270.65,270.65,252.65,180.65,180.6
00104	20*	C 55/,GT(1),GT(2),GT(3),GT(4),GT(5),GT(6),GT(7),GT(8) /-6.5,0.0,1.0,2
00104	21*	C 6.8,0.0,-2.0,-4.0,0.0/P(1),P(2),P(3),P(4),P(5),P(6),P(7),P(8),P(9)
00104	22*	C

```

***** REFP *****
00104 7 /2116.22,472.681,114.345,18.1298,2.31631,1.23225,0.380323,0.02167
00104 82,0.0027138/160,WM0,RU1 /9.80665,28.9644,8314.32,1545.31/CP1,C
00104 9P2,CP3,CP4 /0.239573,-0.000127,0.000051,0.000014/10,0T /270.0,90.
00104 10/,CF1,CF2 /1000.0,0.671969/
00173 DATA A0,A1,A2,A3,A4,A5,A6,A7,A8,A9,A10,A11/
00173 1 0.10E+01, 0.3533367370E-01, -0.7474788290E-03,
00173 2 0.2121572232E-03, -0.1325255219E-04, 0.5344159692E-06,
00173 3 -0.1132274564E-07, 0.1965359762E-09, -0.1723714966E-11,
00173 4 0.8707590786E-14, -0.2341816445E-16, 0.259772972E-19/,
00210 REAL MWT
00211 TMK = REFTP/1.8
00212 REFFVIS = ((1.458E-06*TMK**1.5)/(TMK+110.4))*CF2
00213 REFK = ((6.325E-07*TMK**1.5)/(TMK+245.4+10.0*(-12.0/TMK)))*CF2
00214 REFCP = CPAIR(REFTP)
00215 IF (ELEV.GT.301000.) GO TO 20
00217 ELEV = ELEV*0.3048
00220 GEOM = POLY(7,H,ELEV)
00221 GEOH = GEOM/0.3048
00222 DO 1 I=1,9
00225 DH = GH(I)-GEOHM/CF1
00226 IF (DH.GT.0.0) GO TO 2
00230 1 CONTINUE
00232 2 J = I-1
00233 DH = GH(J)-GEOHM/CF1
00234 TMB = T(J)-GT(J)*DH
00235 IF (J.EQ.2.OR.J.EQ.5.OR.J.EQ.8) GO TO 3
00237 PATM = P(J)*(T(J)/TMB)**(60*WM0*CF1/(RU*GT(J)))
00240 60 TO 4
00241 3 PATM = P(J)*EXP(60*WM0*DH*CF1/(RU*T(J)))
00242 4 REFRHO = WM0*PATM/(RU1*REFTP)
00243 REFR = REFFVIS*REFCP/REFK
00244 REFGAM = REFCP/(REFCP-0.0686)
00245 RETURN
00246 20 Z = ELEV/(3280.8399)
00247 MWT = 28.9644-0.0309491*(Z-90.0)
00250 PZ = 1.0/((A0+Z*(A1+Z*(A2+Z*(A3+Z*(A4+Z*(A5+Z*(A6+Z*(A7+Z
00250 1 *(A8+Z*(A9+Z*(A10+Z*(A11))))))))))**4)
00252 PSL = 2116.22657
00253 PATM = PSL*PZ
00254 REFRHO = PATM*MWT/(1545.31*REFTP)
00255 REFR = REFFVIS*REFCP/REFK
00256 REFGAM = REFCP/(REFCP-(1.98585/MWT))
00257 RETURN
END

```

END OF COMPILATION: NO DIAGNOSTICS.

DHDG.P \*\*\*\*\* RHOF/CFFC43 \*\*\*\*\*

\*\*\*\*\* RHOF/CFFC43 \*\*\*\*\*  
QFOR'S ME\*NASAS5.RHOF/CFFC43,ME\*NASAS5.RHOF/CFFC43  
FOR S9A-07/13/72-20:57:42 (0,)

FUNCTION RHOF ENTRY POINT 000014

STORAGE USED: CODE(1) 000016; DATA(0) 000011; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000004 INJP\$ 0000 R 000000 RHOF 0000 R 000001 X1 0000 R 000002 X2

00101 1\* FUNCTION RHOF(P,T)

00101 2\* C

00101 3\* C

00101 4\* C

00101 5\* C

00101 6\* C

00101 7\* C

00103 8\*

00106 9\*

00107 10\*

00110 11\*

THIS FUNCTION SUBPROGRAM COMPUTES :  
DENSITY AS A FUNCTION OF PRESSURE (LBF/SQ.FT) AND  
TEMPERATURE (R) OF FC-43 UNITS SLUG/GU.FT

DATA X1,X2 /157.0883,-0.076167/  
RHOF = (X1+X2\*T)/32.174  
RETURN  
END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG:P \*\*\*\*\* RHOF/CFFC75 \*\*\*\*\*

\*\*\*\*\* RHOF/CF75 \*\*\*\*\*  
QFOR'S ME=NASAS5.RHOF/CF75;ME=NASAS5.RHOF/CF75  
FOR S9A-07/13/72-20:57:44 (0.)

FUNCTION RHOF ENTRY POINT 000014

STORAGE USED: CODE(1) 000016; DATA(0) 000011; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000004 INJPS 0000 R 000000 RHOF 0000 R 000001 X1 0000 R 000002 X2

00101	1*		FUNCTION RHOF(P,T)
00101	2*	C	
00101	3*	C	THIS FUNCTION SUBPROGRAM COMPUTES :
00101	4*	C	DENSITY AS A FUNCTION OF PRESSURE (LBF/SQ.FT) AND
00101	5*	C	TEMPERATURE (R) OF FC-75
00101	6*	C	UNITS SLUG/QU.FT
00101	7*	C	
00103	8*		DATA X1,X2 /155.522,-0.085/
00106	9*		RHOF = (X1+X2*T)/32.174
00107	10*		RETURN
00110	11*		END

END OF COMPILATION: NO DIAGNOSTICS.

BHDG:P \*\*\*\*\* RHOF/CFHE \*\*\*\*\*

FUNCTION RHOF ENTRY POINT 000067

STORAGE USED: CODE(1) 000077; DATA(0) 000023; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 PF  
0004 CAPPA  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000015 2L 0001 000055 4L 0004 R 000000 CAPPA 0000 R 000005 DIFF 0000 R 000006 DPDRHO  
0000 R 000007 DRHO 0000 000014 INJPS 0000 I 000001 N 0003 R 000000 PF 0000 R 000004 P1  
0000 R 000002 R 0000 R 000000 RHOF 0000 R 000003 RHO1

00101 1\* FUNCTION RHOF(P,T)  
00101 2\* C  
00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
00101 4\* C DENSITY AS A FUNCTION OF PRESSURE (LBF/SQ.FT) AND  
00101 5\* C TEMPERATURE (R) OF HELIUM  
00101 6\* C UNITS SLUG/QU.FT  
00101 7\* C  
00101 8\* N = 1  
00101 9\* R = 12419.164  
00104 10\* RHOF = P/(R\*T)  
00106 11\* IF (0.388-RHO1)1,1,2  
00111 12\* 1 RHOF = 0.388  
00112 13\* 2 P1 = PF(RHO1,T)  
00113 14\* DIFF = ABS((P-P1)/P)  
00114 15\* IF (DIFF-1.0E-12)4,3,3  
00117 16\* 3 DPDRHO = 1.0/(RHO1\*CAPPA(RHO1,T))  
00120 17\* DRHO = (P-P1)/DPDRHO  
00121 18\* RHOF = RHO1+DRHO  
00122 19\* N = N+1  
00123 20\* IF (N-10)2,2,4  
00126 21\* 4 RHOF = RHO1  
00127 22\* RETURN  
00130 23\* END

END OF COMPILATION: NO DIAGNOSTICS.

GH0G\*P \*\*\*\*\* RHOF/CFNAK \*\*\*\*\*

CFOR,S ME\*NASAS.RHOF/CFNAK,ME\*NASAS.RHOF/CFNAK  
 FOR S9A-07/13/72-20:57:49 (0,)

FUNCTION RHOFCFNAK ENTRY POINT 000031

STORAGE USED: CODE(1) 000040; DATA(0) 000017; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 CAPPA  
 0004 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0003 R 000000 CAPPA 0000 000011 INJPS 0000 R 000003 P0 0000 R 000004 RHO  
 0000 R 000000 RHOFCFNAK 0000 R 000005 TK 0000 R 000001 X1 0000 R 000002 X2

00101 1\* FUNCTION RHOFCFNAK  
 00101 2\* C  
 00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES:  
 00101 4\* C DENSITY AS A FUNCTION OF PRESSURE (LBF/SQ.FT) AND  
 00101 5\* C TEMPERATURE (R) OF NAK 78.6  
 00101 6\* C UNITS SLUG/QU.FT  
 00101 7\* C  
 00103 8\* DATA X1,X2 /58.773064,-0.008433/P0 /2116.224/  
 00107 9\* RHO = (X1\*X2\*T)/32.174  
 00110 10\* TK = CAPPA(RHO,T)  
 00111 11\* P1 = P-P0  
 00112 12\* RHOFCFNAK = RHO\*(1.0+TK\*P1)  
 00113 13\* RETURN  
 00114 14\* END

END OF COMPILATION: NO DIAGNOSTICS.

RHOFCFNAK \*\*\*\*\* RHOFCFNAK \*\*\*\*\*

FOR,S ME\*NASAS,RHOF/CFSIL,ME\*NASAS,RHOF/CFSIL  
FOR S9A-07/13/72-20157:51 (0,)

FUNCTION RHOF ENTRY POINT 000122

STORAGE USED: CODE(1) 000127; DATA(0) 000065; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 EXP  
0004 NWDUS  
0005 NIO2S  
0006 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000	000027	1F	0000	R	000022	A	0000	R	000001	A1	0000	R	000002	A2	0000	R	000003	A3	
0000	R	000004	A4	0000	R	000005	A5	0000	R	000023	B	0000	R	000006	B1	0000	R	000007	B2
0000	R	000010	B3	0000	R	000011	B4	0000	R	000012	B5	0000	R	000024	C	0000	R	000013	C1
0000	R	000014	C2	0000	R	000017	DT	0000	000054	INJPS	0000	R	000025	P1	0000	R	000026	RHO	
0000	R	000000	RHOF	0000	R	000020	THETA	0000	R	000021	THETA1	0000	R	000015	T0	0000	R	000016	T01

00101 1\* FUNCTION RHOF(P,T)

00101 2\* C  
 00101 3\* C  
 00101 4\* C  
 00101 5\* C  
 00101 6\* C  
 00101 7\* C  
 00101 8\* C  
 00103 9\*  
 00103 10\*  
 00103 11\*  
 00123 12\*  
 00124 13\*  
 00125 14\*  
 00126 15\*  
 00127 16\*  
 00130 17\*  
 00131 18\*  
 00132 19\*  
 00133 20\*  
 00135 21\*  
 00140 22\*  
 00140 23\*  
 00141 24\*  
 00142 25\*

THIS FUNCTION SUBPROGRAM COMPUTES:  
 DENSITY AS A FUNCTION OF PRESSURE (LBF/50.FT) AND  
 TEMPERATURE (R) OF DOW CORNING 200 SILICON OIL (1 CS)  
 UNITS SLUG/CU.FT  
 TEMPERATURE .GE. 359.67 AND .LE. 859.67

DATA A1,A2,A3,A4,A5 /12.35,2.98333,1.1,-0.48333,0.1/,B1,B2,B3,B4,  
 185 /-1.5,-0.01333,-1.18,0.57333,-0.1/,C1,C2 /0.7767,-0.0288/,  
 210,T01,DT /559.67,609.67,50.0/  
 THETA = (T-T01)/DT  
 THETA1 = (T-T01)/DT  
 A = (((A5\*THETA+A4)\*THETA+A3)\*THETA+A2)\*THETA+A1)\*1.0E-06  
 B = (((B5\*THETA+B4)\*THETA+B3)\*THETA+B2)\*THETA+B1)\*1.0E-09  
 C = C1+C2\*THETA1  
 P1 = P/144.0-14.696  
 RHO = C\*EXP(A\*P1+0.5\*B\*P1\*P1)  
 RHOF = RHO\*62.42/32.174  
 IF (T.GE.360.67.AND.T.LE.860.67) RETURN  
 WRITE (6,1) T  
 1 FORMAT (1H0,4HDENSITY OF SILICON OIL, TEMP. OUT OF RANGE. T = ,F8  
 1.3,/) T  
 RETURN  
 END

END OF COMPILATION: NO DIAGNOSTICS.

\*\*\*\*\* RHOF/CFSIL \*\*\*\*\*  
DHOG,P \*\*\*\*\* RKS \*\*\*\*\*

DATE 071372

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\*\*\*\*\* RKS \*\*\*\*\*

FOR,S ME,NASAS,RKS,ME,NASAS,RKS  
FOR S9A-07/13/72-20:57:54 (0,)

SUBROUTINE RKS ENTRY POINT 000643

STORAGE USED: CODE(1) 001040; DATA(0) 000064; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR2\$  
0004 NEXP5\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000010	10L	0001	000313	110L	0001	000333	120L	0001	000045	126G	0001	000343	130L	
0001	000071	140G	0001	000355	140L	0001	000105	146G	0001	000130	156G	0001	000417	160L	
0001	000150	164G	0001	000177	174G	0001	000500	185L	0001	000510	190L	0001	000013	20L	
0001	000232	205G	0001	000270	217G	0001	000524	220L	0001	000530	230L	0001	000543	240L	
0001	000374	243G	0001	000032	25L	0001	000552	250L	0001	000554	251L	0001	000572	257L	
0001	000604	259L	0001	000623	270L	0001	000456	300L	0001	000615	336G	0001	000054	40L	
0000	R	000014	AM	0001	000006	5L	0001	000076	50L	0001	000123	70L	0001	000135	80L
0000	R	000014	AM	0000	R	000007	AMAX	0000	R	000011	C	0000	R	000001	DDT
0000	R	000003	DELT	0000	R	000012	E	0000	R	000000	FR10	0000	I	000005	IFLAG
0000	000030	INJP\$	0000	I	000002	ISYMP	0000	I	000013	J	0000	R	000006	S	

00101	1*	SUBROUTINE RKS(deriv,ctrl,y,dy,a,r,t,del,n,ifvd,ibkp,ntry,	D6000100
00101	2*	11err,dely,pd,sd,ys,yt,dyst,ysimp)	D6000200
00103	3*	dimension y(n),dy(n),a(n),r(n),dely(n),	D6000300
00103	4*	ipd(n),sd(n),ys(n),dyst(n),yst(n),ysimp(n)	D6000400
00104	5*	external deriv, ctrl	D6000500
00104	6*	fr10 is fifth root of ten	D6000600
00105	7*	fr10=1.5848932	D6000700
00106	8*	1err=0	D6000800
00106	9*		
00106	10*	THIS SUBROUTINE COMPUTES :	
00106	11*	RUNGE-KUTTA INTEGRATION (PRIMARY)	
00106	12*		
00106	13*	YS CONTAINS Y VALUES AT LEFT END POINT OF INTEGRATION INTERVAL	D6000900
00106	14*		D6001000
00106	15*	YSIMP CONTAINS Y FOR SIMPSONS RULE CHECK CHECK NOT MADE FOR	D6001100
00106	16*	FIXED STEP MODE ISYMP IS CONTROL PARAMETER =1, FIXED, 2 VARDG	D6001200
00106	17*		D6001300
00106	18*	IF FIXED STEP SIZE GO ONE INTERVAL OF LENGTH DELT AND RETURN TO	D6001400
00106	19*	CTRL, IF VAR GO TWO INTERVALS BEFORE RETURN TO CTRL	D6001500
00106	20*		D6001600
00106	21*	IFVD = 0 VARIABLE INTERVAL	D6001700
00106	22*	= 1 FIXED	D6001800
00106	23*	IBKP = 0 CUT INTERVAL ONCE BEFORE REPEAT (UNDER IFVD=0)	D6001900
00106	24*	= 1 CUT AS REQUIRED	D6002000

\*\*\*\*\* RKS \*\*\*\*\*

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00106 25* C NTRY = 1 CONTINUE INTEGRATING
00106 26* C RETURN FROM RKS
00106 27* C 2 STEP REPEATED WITH NEW DELT
00106 28* C 3 RESTART
00106 29* C 4
00106 30* C IERR = 0 NORMAL
00106 31* C -1 DELT=0, RETURN FROM RKS
00106 32* C 1 A(I)+ R(I)*ABS(Y(I)) = 0. , RETURN FROM RKS
00107 5 IF(DEL) 20,10,20
00112 10 IERR=-1
00113 60 TO 270
00114 20 CALL DERIV(Y,DY,T)
00115 NTRY=1
00116 CALL CNTRL(Y,DY,DEL,T,NTRY,IFVD)
00117 25 DDT=DEL
00120 IF(IFVD) 40,30,40
00123 30 ISYMP=2
00124 DELT=DEL/2.
00125 DO 31 I=1,N
00130 31 SD(I)=0.0
00132 IFLAG=1
00133 S=1.
00134 60 TO 45
00135 40 ISYMP=1
00136 DELT=DEL
00137 45 DO 46 I=1,N
00142 46 YST(I)=Y(I)
00143 46 DYST(I)=DY(I)
00145 50 DO 60 I=1,N
00150 53* DELY(I)=DELT*DY(I)
00151 54* PD(I)=DELY(I)
00152 55* CONTINUE
00154 60 TO (80,70), ISYMP
00155 70 DO 71 I=1,N
00160 71 SD(I)=SD(I)+S*DY(I)
00162 59* T=T+DELT/2.
00163 60* YS(I)=Y(I)
00166 61* Y(I)=YS(I)
00167 62* Y(I)=YS(I)+DELY(I)/2.
00170 63* CONTINUE
00172 85 CALL DERIV(Y,DY,T)
00173 DO 90 I=1,N
00176 65* DELY(I)=DELT*DY(I)
00177 66* PD(I)=PD(I)+2.*DELY(I)
00177 67* Y(I)=YS(I)+DELY(I)/2.
00200 68* CONTINUE
00201 90 CALL DERIV(Y,DY,T)
00203 70* DO 95 I=1,N
00204 71* DELY(I)=DELT*DY(I)
00207 72* PD(I)=PD(I)+2.*DELY(I)
00210 73* Y(I)=YS(I)+DELY(I)
00211 74* CONTINUE
00212 75* T=T+DELT/2.
00214 76* CALL DERIV(Y,DY,T)
00215 77* DO 100 I=1,N
00216 78* DELY(I)=DELT*DY(I)
00221 79* PD(I)=PD(I)+DELY(I)
00222 80* Y(I)=YS(I)+PD(I)/6.
00223 81*

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00224 82* 100 CONTINUE
00226 83* GO TO (110,120),ISYMP
00227 84* 110 NTRY=1
00230 85* CALL DERIV(Y,DY,T)
00231 86* CALL CNTRL(Y,DY,DEL,T,NTRY,IFVD)
00232 87* GO TO 300
00233 88* 120 GO TO (130,140),IFLAG
00234 89* 130 S=4.
00235 90* IFLAG=2
00236 91* CALL DERIV(Y,DY,T)
00237 92* GO TO 50
00240 93* 140 CALL DERIV(Y,DY,T)
00241 94* DO 180 I=1,N
00242 95* SD(I)=SD(I)+DY(I)
00245 96*
00246 97* YSYMP(I)=YST(I)+DEL*SD(I)/3.
00247 98* D =ABS(Y(I))-YSYMP(I)
00250 99* C =A(I)+R(I)*ABS(Y(I))
00251 100* IF(C ) 160,150,160
00254 101* 150 IERR=1
00255 102* GO TO 270
00256 103* E =ABS(D /C )
00257 104* AMAX=AMAX1(AMAX,E)
00260 105* 160 CONTINUE
00262 106* IF(AMAX-1.) 215,215,230
00265 107* 215 NTRY= 1
00266 108* CALL CNTRL(Y,DY,DEL,T,NTRY,IFVD)
00267 109* 300 IF(NTRY-1) 185,185,310
00272 110* 310 IF(NTRY-2) 270,270,330
00275 111* 330 IF(NTRY-3) 340,340,5
00300 112* 340 T=T-DO1
00301 113* IF(DEL) 259,10,259
00304 114* 185 GO TO (40,190),ISYMP
00305 115* 190 IF(AMAX-.75) 200,25,220
00310 116* 200 IF(AMAX-.075) 210,25,25
00313 117* 210 DEL=DEL*FR10
00314 118* GO TO 25
00315 119* 220 DEL=DEL/FR10
00316 120* GO TO 25
00317 121* 230 I =I+ IBXP
00320 122* 60 GO TO (240,250),I
00321 123* 240 T=T-DEL
00322 124* DEL=DEL/FR10
00323 125* GO TO 259
00324 126* 250 J=1
00325 127* 251 AM=AMAX/10.**J
00326 128* IF(1.-AM) 255,257,257
00331 129* 255 J=J+1
00332 130* GO TO 251
00333 131* 257 T=T-DEL
00334 132* DEL=DEL/(FR10**J)
00335 133* 259 DO 245 I=1,N
00340 134* DY(I)=DYST(I)
00341 135* 245 Y(I)=YST(I)
00343 136* GO TO 25
00344 137* 270 RETURN
00345 138* END

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\*\*\*\*\* RMS \*\*\*\*\*

END OF COMPILATION: NO DIAGNOSTICS.

DHDS.P \*\*\*\*\* RKSF \*\*\*\*\*

FOR S ME\*NASAS5,RKSF,ME\*NASAS5,RKSF  
FOR S9A-07/13/72-20:57:59 (0,)

SUBROUTINE RKSF ENTRY POINT 000643

STORAGE USED: CODE(1) 001040/ DATA(0) 000064/ BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR2\$  
0004 NEXP5\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000010	10L	0001	000313	110L	0001	000333	120L	0001	000045	126G	0001	000343	130L
0001	000071	140G	0001	000355	140L	0001	000105	146G	0001	000130	156G	0001	000417	160L
0001	000150	164G	0001	000177	174G	0001	000500	185L	0001	000510	190L	0001	000013	20L
0001	000232	205G	0001	000270	217G	0001	000524	220L	0001	000530	230L	0001	000543	240L
0001	000374	243G	0001	000032	25L	0001	000552	250L	0001	000554	251L	0001	000572	257L
0001	000604	259L	0001	000623	270L	0001	000456	300L	0001	000615	336G	0001	000054	40L
0001	000060	45L	0001	000006	5L	0001	000076	50L	0001	000123	70L	0001	000135	80L
0000 R	000014	AM	0000 R	000007	AMAX	0000 R	000011	C	0000 R	000010	D	0000 R	000001	DDT
0000 R	000003	DELT	0000 R	000012	E	0000 R	000000	FR10	0000 I	000004	I	0000 I	000005	IFLAG
0000	000030	INJPS	0000 I	000002	ISYMP	0000 I	000013	J	0000 R	000006	S			

00101	1*	SUBROUTINE RKSF(DERIV,CNTRL,Y,DY,A,R,T,DEL,N,IFVD,IBKP,NTRY,	D6000100
00101	2*	IERR,DELY,PD,SD,YS,YST,DYST,YSIMP)	D6000200
00103	3*	DIMENSION Y(N),DY(N),A(N),R(N),DELY(N),	D6000300
00103	4*	IPD(N),SD(N),YS(N),DYST(N),YST(N),YSIMP(N)	D6000400
00104	5*	EXTERNAL DERIV, CNTRL	D6000500
00104	6*	FR10 IS FIFTH ROOT OF TEN	D6000600
00105	7*	FR10=1.5848932	D6000700
00106	8*	IERR=0	D6000800
00106	9*		
00106	10*	THIS SUBROUTINE COMPUTES :	
00106	11*	RUNGE-KUTTA INTEGRATION (SECONDARY)	
00106	12*	YS CONTAINS Y VALUES AT LEFT END POINT OF INTEGRATION INTERVAL	D6000900
00106	13*		D6001000
00106	14*	YSIMP CONTAINS Y FOR SIMPSONS RULE CHECK CHECK NOT MADE FOR	D6001100
00106	15*	FIXED STEP MODE ISYMP IS CONTROL PARAMETER =1, FIXED, 2 VARD	D6001200
00106	16*		D6001300
00106	17*	IF FIXED STEP SIZE GO ONE INTERVAL OF LENGTH DELT AND RETURN TO	D6001400
00106	18*	CNTRL, IF VAR GO TWO INTERVALS BEFORE RETURN TO CNTRL	D6001500
00106	19*		D6001600
00106	20*	IFVD = 0 VARIABLE INTERVAL	D6001700
00106	21*	IBKP = 1 FIXED	D6001800
00106	22*	CUT INTERVAL ONCE BEFORE REPEAT (UNDER IFVD=0)	D6001900
00106	23*	IBKP = 0 CUT AS REQUIRED	D6002000
00106	24*	NTRY = 1 CONTINUE INTEGRATING	D6002100

```

*****
00106 25* RKSF *****
00106 26* C 2 RETURN FROM RKS
00106 27* C 3 STEP REPEATED WITH NEW DELT
00106 28* C 4 RESTART
00106 29* C IERR = 0 NORMAL
00106 30* C -1 DELT=0, RETURN FROM RKS
00107 31* C 1 A(I)+R(I)*ABS(Y(I)) = 0, RETURN FROM RKS
00112 5 IF(DEL) 20,10,20
00112 10 IERR=-1
00113 30 TO 270
00114 20 CALL DERIV(Y,DY,T)
00115 35* NTRY=1
00116 36* CALL CNTRL(Y,DY,DEL,T,NTRY,IFVD)
00117 37* 25 DT=DEL
00120 38* IF(IFVD) 40,30,40
00123 39* 30 ISYMP=2
00124 40* DELT=DEL/2.
00125 41* DO 31 I=1,N
00130 42* 31 SD(I)=0.0
00132 43* IFLAG=1
00133 44* S=1.
00134 45* GO TO 45
00135 46* 40 ISYMP=1
00136 47* DELT=DEL
00137 48* 45 DO 46 I=1,N
00142 49* Y(I)=Y(I)
00143 50* 46 DYST(I)=DY(I)
00145 51* 50 DO 60 I=1,N
00150 52* DELY(I)=DELT*DY(I)
00151 53* PD(I)=DELY(I)
00152 54* 60 CONTINUE
00154 55* GO TO (80,70),ISYMP
00155 56* 70 DO 71 I=1,N
00160 57* 71 SD(I)=SD(I)+S*DY(I)
00162 58* 80 T=T+DELT/2.
00163 59* DO 85 I=1,N
00166 60* Y(I)=Y(I)
00167 61* Y(I)=Y(I)+DELY(I)/2.
00170 62* 85 CONTINUE
00172 63* CALL DERIV(Y,DY,T)
00173 64* DO 90 I=1,N
00176 65* DELY(I)=DELT*DY(I)
00177 66* PD(I)=PD(I)+2.*DELY(I)
00200 67* Y(I)=Y(I)+DELY(I)/2.
00201 68* 90 CONTINUE
00203 69* CALL DERIV(Y,DY,T)
00204 70* DO 95 I=1,N
00207 71* DELY(I)=DELT*DY(I)
00210 72* PD(I)=PD(I)+2.*DELY(I)
00211 73* Y(I)=Y(I)+DELY(I)
00212 74* 95 CONTINUE
00214 75* T=T+DELT/2.
00215 76* CALL DERIV(Y,DY,T)
00216 77* DO 100 I=1,N
00221 78* DELY(I)=DELT*DY(I)
00222 79* PD(I)=PD(I)+DELY(I)
00223 80* Y(I)=Y(I)+PD(I)/6.
00224 81* 100 CONTINUE

```

DG002200  
 DG002300  
 DG002400  
 DG002500  
 DG002600  
 DG002700  
 DG002800  
 DG002900  
 DG003000  
 DG003100  
 DG003200  
 DG003300  
 DG003400  
 DG003500  
 DG003600  
 DG003700  
 DG003800  
 DG003900  
 DG004000  
 DG004100  
 DG004200  
 DG004300  
 DG004400  
 DG004500  
 DG004600  
 DG004700  
 DG004800  
 DG004900  
 DG005000  
 DG005100  
 DG005200  
 DG005300  
 DG005400  
 DG005500  
 DG005600  
 DG005700  
 DG005800  
 DG005900  
 DG006000  
 DG006100  
 DG006200  
 DG006300  
 DG006400  
 DG006500  
 DG006600  
 DG006700  
 DG006800  
 DG006900  
 DG007000  
 DG007100  
 DG007200  
 DG007300  
 DG007400  
 DG007500  
 DG007600  
 DG007700  
 DG007800

```

00226 82* GO TO (110,120),ISYMP
00227 83* 110 NTRY=1
00230 84* CALL DERIV(Y,DY,T)
00231 85* CALL CNTRL(Y,DY,DEL,T,NTRY,IFVO)
00232 86* GO TO 300
00233 87* 120 GO TO (130,140),IFLAG
00234 88* 130 S=4.
00235 89* IFLAG=2
00236 90* CALL DERIV(Y,DY,T)
00237 91* GO TO 50
00240 92* 140 CALL DERIV(Y,DY,T)
00241 93* AMAX =0.0
00242 94* DO 180 I=1,N
00245 95* SD(I)=SD(I)+DY(I)
00246 96* YSIMP(I)=YST(I)+DEL*SD(I)/3.
00247 97* D =ABS(Y(I))-YSIMP(I)
00250 98* C =A(I)+R(I)*ABS(Y(I))
00251 99* IF(C ) 160,150,160
00254 100* 15) IERR=1
00255 101* GO TO 270
00256 102* 16) E =ABS(D /C )
00257 103* AMAX=AMAX1(AMAX,E)
00260 104* 18) CONTINUE
00262 105* IF(AMAX<1.) 215,215,230
00265 106* 215 NTRY= 1
00266 107* CALL CNTRL(Y,DY,DEL,T,NTRY,IFVO)
00267 108* 30) IF(NTRY-1) 185,185,310
00272 109* 31) IF(NTRY-2) 270,270,330
00275 110* 33) IF(NTRY-3) 340,340,5
00300 111* 34) T=T-DDT
00301 112* IF(DEL) 259,10,259
00304 113* 185 GO TO (40,190),ISYMP
00305 114* 19) IF(AMAX<-.75) 200,25,220
00310 115* 20) IF(AMAX<-.075) 210,25,25
00313 116* 21) DEL=DEL*FR10
00314 117* GO TO 25
00315 118* 22) DEL=DEL/FR10
00316 119* GO TO 25
00317 120* 23) I =I+ 18KP
00320 121* GO TO (240,250),I
00321 122* 24) T=T-DEL
00322 123* DEL=DEL/FR10
00323 124* GO TO 259
00324 125* 250 J=1
00325 126* 251 AM=AMAX/10.**J
00326 127* IF(1.-AM) 255,257,257
00331 128* 255 J=J+1
00332 129* GO TO 251
00333 130* 257 T=T-DEL
00334 131* DEL=DEL/(FR10**J)
00335 132* 259 DO 245 I=1,N
00340 133* DY(I)=YST(I)
00341 134* 245 Y(I)=YST(I)
00343 135* GO TO 25
00344 136* 270 RETURN
00345 137* END

```

\*\*\*\*\* RMSF \*\*\*\*\*

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* SHADE \*\*\*\*\*

DATE 071372

PAGE

4



\*\*\*\*\* SHADE \*\*\*\*\*

QFOR,S ME\*NASAS.SHADE,ME\*NASAS.SHADE  
FOR 59A-07/13/72-20:58:04 (0,)

SUBROUTINE SHADE ENTRY POINT 000073

STORAGE USED: CODE(1) 000104; DATA(0) 000023; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000027 10L 0001 000015 1066 0001 000057 1236 0000 000005 INJPS 0000 I 000001 J  
0000 I 000003 J1 0000 I 000002 J2 0000 R 000000 QM

```

00101 1* SUBROUTINE SHADE(Q,NS)
00103 2* DIMENSION Q(1)
00104 3* QM = 0.0
00105 4* DO 10 J = 1,NS
00110 5* IF (QM.GT.Q(J)) GO TO 10
00112 6* QM = Q(J)
00113 7* J2 = J
00114 8* CONTINUE
00116 9* J1 = 1
00117 10* IF (J2.LE.NS/2) J1 = NS/2+1
00121 11* J2 = J1+NS/2-1
00122 12* DO 20 J = J1,J2
00125 13* Q(J) = 0.0
00127 14* RETURN
00130 15* END

```

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* SHAPEF \*\*\*\*\*

\*\*\*\*\* SHAPE \*\*\*\*\*  
 FOR S ME\*NASAS.SHAPE, ME\*NASAS.SHAPE  
 FOR S9A-07/13/72-20:58:06 (07)

SUBROUTINE SHAPE ENTRY POINT 001020

STORAGE USED: CODE(1) 001035; DATA(0) 000152; BLANK COMMON(2) 000000

COMMON BLOCKS:

0003 GRD 003721  
 0004 SSF 000010

EXTERNAL REFERENCES (BLOCK, NAME)

0005 ACOS  
 0006 ATAN  
 0007 SQR  
 0010 COS  
 0011 SIN  
 0012 NEXP65  
 0013 ASIN  
 0014 ALOG  
 0015 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000036	1179	0001	000037	1226	0001	000040	1256	0001	000066	1356	0001	000307	1746									
0001	000053	20L	0001	000447	2166	0001	000452	2216	0001	000457	2256	0001	000671	2616									
0001	000752	2746	0001	000107	30L	0001	000757	3006	0001	000115	35L	0001	000603	62L									
0000	R	000620	64L	0001	000634	65L	0000	R	000046	A	0000	R	000002	ALPFN									
0000	R	000047	AR	0000	R	000050	ARC	0000	R	000051	AR2	0000	R	000027	BETA								
0000	R	000066	BETA1	0000	R	000067	BETA2	0000	R	000014	CF	0000	R	000465	CSF								
0000	R	000015	DCL	0003	003464	DX2	0003	003472	DX21	0000	R	000064	DZ	0003	R	003463	OZMFN						
0000	R	000045	DZ2	0003	003410	EBFN	0003	003441	EBMP	0003	003467	EXTIFN	0003	003471	EXTIMP	0000	I	000021	I				
0003	003466	EXT3FN	0003	003470	EXTSMP	0000	R	000062	GAMA	0000	R	000036	GAMA	0000	I	000024	L	0003	R	003461	LCT		
0000	000112	INJPS	0000	I	000022	J	0000	I	000023	K	0000	R	000032	OB	0000	R	000034	OP	0000	R	000012	PI	
0000	R	000033	PB	0000	R	000017	PHI	0000	R	000037	PHIO	0000	R	000055	PHI1	0000	R	000030	RHO2	0000	R	000042	RR02
0000	R	000013	PI2	0000	R	000040	PR4	0000	R	000044	RC	0000	R	000031	RHO	0000	R	000042	RR02	0004	R	000000	R2
0000	R	000053	RL	0000	R	000057	RL2	0000	R	000016	RR	0004	R	000003	R1	0004	R	000001	T	0000	R	000020	THETA
0000	R	000054	RT	0000	R	000061	RTL	0003	R	003670	SST	0004	R	000006	WVA	0004	R	000007	WWB	0000	R	000005	XX
0000	R	000060	THETA1	0003	000000	TRMTX	0000	R	000070	WF	0004	R	000004	XSPM	0003	R	003453	ZZ2					
0000	R	000000	X	0000	R	000065	XB	0000	R	000025	Y												
0003	R	003446	XX2	0000	R	000025	Y																

00101 1\* C SUBROUTINE SHAPE  
 00101 2\* C  
 00101 3\* C THIS SUBROUTINE COMPUTES THE SHAPE FACTOR BETWEEN :  
 00101 4\* C 1- TUBE AND NEAR FIN

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00101 5* C 1- TUBE AND FAR FIN
00101 6* C 3- ADJACENT TUBES
00101 7* C
00103 8* COMMON /GRD/ TRMIX(30*60),EBEN(5,5),EBMP(5),XX2(5),ZZ2(5),
00103 9* MCVRD,LCI,LLI,DZMFN,
1 DXX2,CSF,EXTSFN,EXTIFN,EXTSMP,EXTIMP,DXX21,SS(5,5,5)
2 ,SS1(5,5)
3 COMMON /SSF/R2, I, ALPFN,R1,XSP#,XSMP,WWA,WWB
COMMON /DIMENSION X(5), XX(5)
PI = 3.14159
PI2 = PI/2.0
CF = DZMFN*4.0
DCL = 2.0+2.0*R1
RR = R2*R2
PHI = ACOS((XSMP+R1)/R2)
THETA = 1.570796-PI
CSF = 1.0/(R2*THETA)
DO 10 I=1,5
DO 10 J=1,5
DO 10 K=1,5
DO 10 SS(I,J,K) = 0.0
L = 0
20 DO 50 I=1,5
IF (L.EQ.1) GO TO 30
X(I) = XX2(I)+R1
XX(I) = XX2(I)
Y = WWA*X(I)+WWB
YY = Y*Y
IF (L.EQ.0) GO TO 35
30 X(I) = DCL-X(I)
XX(I) = 2.0-XX2(I)
35 BETA = ATN(Y/X(I))
RHO = X(I)*X(I)+YY
RHO = SQRT(RHO2)
O3 = ACOS(R2/RHO)
P3 = PHI-BETA
OP = OB-PB
BA = BETA+ALPFN
GAMMA = OB+BA
IF (L.EQ.1) BA = BETA-ALPFN
PHIO = ATN(DZMFN/(2.0*XX(I)))
SS(I,1,1) = ((1.0-SIN(GAMMA))/PI)*(PHIO*SIN(2.0*PHIO)/2.0)*4.0+
1 SS(I,1,1)
PR4 = 4.*PI*RHO
KRO = 2.*R2*RHO
KRO2 = RRO*RRO
CPB = COS(PB)
KC = RRO*CPB
J = 1
DO 50 K=2,5
DZ2 = ((ZZ2(K)-ZZ2(J))*CF)**2
A = RR+RHO2+DZ2
AR = A*A-RR02
ARC = A-RC
AR2 = A-2.0*RR
SS(I,J,K) = (COS(BA)/PR4*(ALOG(ARC/AR2)-1.0+AR2/ARC)+SIN(BA)/PR4*
1 (OP+2.0*RR02+DZ2-A*AR)/(AR**1.5))*(ASIN((RRO-A*COS(OB)

```

```

***** SHAPE *****
02003 2 )/AR2)-ASIN((RR0-A*CPB)/ARC))+(4.0*A*DZ2/AR-2.0)*((R2
02004 3 *RHO*SIN(OB))/AR2-(R2*RHO*SIN(PB))/ARC)))*CF+SS(I,J,K)
02006 50 CONTINUE .LT. 0.0) SS(I,J,K) = 0.0
02011 IF (L.EG,1) GO TO 70
02013 L = 1
02014 60 TO 20
02015 70 DO 60 I=1,5
02020 80 DO 60 K=2,5
02023 SS(I,K,1) = SS(I,1,K)
02024 DO 60 J=2,5
02027 M = ABS(J-K)+1
02030 60 SS(I,J,K) = SS(I,1,M)
02034 RL = DCL/R2-COS(PHI)
02035 RT = T/R2
02036 PHI1 = ATAN(RT/RL)
02037 RT2 = RT*RT
02040 RL2 = RL*RL
02041 THETA1 = ACOS(1.0/SQRT(RT2+RL2))
02042 KTL = (1.0-RT2)/RL2
02043 GAMA = PI2-PHI1-THETA1
02044 IF (RTL-1.0E-03) 62,62,61
02047 61 SST = (GAMA+RL*(SQRT(1.0-RTL)-1.0))/(PI2-PHI)
02050 60 TO 65
02051 62 IF(RTL-1.0E-04) 63,64,64
02054 63 SST = (GAMA-RL*RTL/2.0)/(PI2-PHI)
02055 60 TO 65
02056 64 SST = (GAMA-RL*(RTL/2.0+RTL*RTL/8.0))/(PI2-PHI)
02057 65 J = 1
02060 DO 80 K=1,5
02063 91* = (ZZ2(K)-ZZ2(J))*CF
02064 92* = 2.0*(1.0*R1-R2)*COS((PHI1+THETA1)/2.0)
02065 93* = ATAN((DZ-DZMFN/2.0)/XB)
02066 94* = ATAN((DZ+DZMFN/2.0)/XB)
02067 95* = (BETA2-BETA1)/2.0*(SIN(2.0*BETA2)-SIN(2.0*BETA1))/4.0
02070 96* = BETA2-BETA1/2.0*(SIN(2.0*BETA2)-SIN(2.0*BETA1))/4.0
02071 97* = 8.0*SST*W/Pi
02072 98* CONTINUE
02073 80 CONTINUE
02074 DO 90 K=2,5
02075 99* = SSS(I,K)
02076 100* = SSS(K,1) = SSS(1,K)
02077 101* DO 90 J=2,5
02078 102* M = ABS(J-K)+1
02079 103* = SSS(J,K) = SSS(1,M)
02080 104* RETURN
02081 105* END

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END OF COMPILATION: NO DIAGNOSTICS.

\*\*\*\*\* TCALC \*\*\*\*\*

\*\*\*\*\* TCALC \*\*\*\*\*

QFOR,S ME\*NASAS,TCALC,ME\*NASAS,TCALC  
FOR S9A-07/13/72-20:58:10 (07)

SUBROUTINE TCALC ENTRY POINT 000635

STORAGE USED: CODE(1) 000652; DATA(0) 000670; BLANK COMMON(2) 000000

COMMON BLOCKS:

0003 ADBH 000352  
0004 TC 000042  
0005 QIN 001610

EXTERNAL REFERENCES (BLOCK, NAME)

0006 YINT  
0007 SHAJE  
0010 NRBJS  
0011 NI02\$  
0012 NBS\$  
0013 NI01\$  
0014 NEXP6\$  
0015 NREW\$  
0016 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000024 10L 0001 000426 110L 0001 000024 125G  
0001 000127 161G 0001 000146 173G 0001 000164 203G  
0001 000276 231G 0001 000504 261G 0001 000042 30L  
0004 R 000003 ABE\*MF 0003 000001 AL 0000 R 000362 ANG  
0003 000147 CP 0003 000000 DTL 0003 000326 DTL  
0003 000013 H 0000 R 000623 HA 0003 000231 HAR  
0000 I 000011 I 0004 I 000001 ICASE 0000 000637 INJPS  
0000 I 000615 IU 0000 I 000613 IUL 0000 I 000612 J  
0000 I 000621 K 0003 R 000064 M 0000 I 000610 NS  
0005 I 001604 NTN 0000 R 000620 PANG 0004 R 000004 PHIN  
0000 R 000124 GALB 0005 000454 G1FN 0005 001274 GITB  
0000 R 000244 QPLN 0000 R 000364 QS 0005 000144 GSFN  
0005 001606 QTO 0000 R 000607 STFBOL 0003 000217 TB  
0003 000040 TIN 0003 000270 TL 0005 000000 TM  
0003 R 000077 TSTAR 0005 001607 TX 0003 000135 VSC

00101 SUBROUTINE TCALC(NTF)  
00103 PARAMETER NTP = 10  
00104 PARAMETER MAXSID = 20  
00105 PARAMETER NTP1 = NTP+1  
00106 PARAMETER MAXSD1 = MAXSID+1  
00107 REAL M  
00110 COMMON /ADBH/ D,AL(NTP),H(NTP1),TH(NTP),TIN(NTP),PIN(NTP),

\*\*\*\*\* TCALC \*\*\*\*\*

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00110 8* 1 M(NTP),NT,STAR(NTP,3),VSC(NTP),CP(NTP),CONFL(NTP),
00110 9* 2 CONFNA(NTP),EMIS(NTP),TB(NTP),HAB(NTP),EFF(NTP,2),TL(NTP,2),
00110 10* 3 GOUT(NTP),DTL(NTP),TOUT(NTP)
00111 11* COMMON /TC/ ITAPE,ICASE,TO,ABEMF,PHIN(NTP,3)
00112 12* COMMON /GIN/ TM(100),GSFN(100,2),GIFN(100,2),GSTB(100,2),
00113 13* QITB(100,2),NTM,NSRD,GTO,IX
00114 14* DIMENSION TIM(4),GSUN(4,MAXSID),GALB(4,MAXSID),GPLN(4,MAXSID),
00115 15* GS(MAXSD1,2),GA(MAXSD1,2),OP(MAXSD1,2),ANG(MAXSD1)
00116 16* DATA SIFBOL/0.1714E-8/
00117 17* READ (ITAPE) NS,TIM(1)
00122 18* IF (ICASE.LE.1) GO TO 30
00124 19* DO 20 I = 2,ICASE
00127 20* TIM(2) = TIM(1)
00130 21* READ (ITAPE) NS,TIM(1)
00134 22* IF (TIM(1).GT.TIM(2)) GO TO 10
00136 23* CONTINUE
00140 24* DO 40 J = 0,NS
00143 25* ANG(J+1) = FLOAT(J*360+180)/NS
00145 26* READ (ITAPE)
00147 27* I = 2
00150 28* READ (ITAPE) NS,TIM(1)
00154 29* I = I+1
00155 30* IF (TIM(1).LT.TO.AND.I.LT.NTM-1) GO TO 50
00157 31* DO 60 I = 1,3
00162 32* BACKSPACE ITAPE
00164 33* DO 70 I = 1,4
00167 34* READ (ITAPE) NS,TIM(1),(GSUN(I,J),GALB(I,J),GPLN(I,J),J=1,NS)
00202 35* DO 80 J = 1,NS
00205 36* GS(J,1) = YINT(TIM,GSUN(I,J),4,4,TO)
00206 37* GA(J,1) = YINT(TIM,GALB(I,J),4,4,TO)
00207 38* GP(J,1) = YINT(TIM,GPLN(I,J),4,4,TO)
00211 39* GS(NS+1,1) = GS(1,1)
00212 40* GA(NS+1,1) = GA(1,1)
00213 41* GP(NS+1,1) = GP(1,1)
00214 42* IUL = 1
00215 43* IF (ISYM.EQ.3) IUL = 3
00217 44* DO 120 IU = 1,IUL
00222 45* DO 120 I = 1,NTF
00225 46* J1 = 0
00226 47* J1 = J1+1
00227 48* IF (ANG(J1)+ANG(1).LE.PHIN(I,IU)) GO TO 90
00231 49* DO 100 J = 1,NS
00234 50* J2 = MIN(MAX(MOD(J+J1-2,NS),1),NS-2)
00235 51* PANG = ANG(J)+PHIN(I,IU)
00236 52* IF (PANG.GE.ANG(NS+1)) PANG = PANG-360.0
00240 53* GS(J,2) = YINT(ANG(J2),GS(J2,1),4,4,PANG)
00241 54* GA(J,2) = YINT(ANG(J2),GA(J2,1),4,4,PANG)
00242 55* GP(J,2) = YINT(ANG(J2),GP(J2,1),4,4,PANG)
00244 56* CALL SHADE(GS(1,2),NS)
00245 57* CALL SHADE(GA(1,2),NS)
00246 58* CALL SHADE(GP(1,2),NS)
00247 59* PANG = AMOD(PHIN(I,IU)+90.0,360.0)
00250 60* J1 = 0
00251 61* J1 = J1+1
00252 62* IF (ANG(J1).LE.PANG) GO TO 110
00254 63* J2 = 0
00255 64* IF (PANG.GE.180.0) J2 = NS/2

```

\*\*\*\*\* TCALC \*\*\*\*\*

```

00257 65*
00260 66*
00263 67*
00265 68*
00266 69*
00267 70*
00270 71*      120
00274 72*
00275 73*
00276 74*

```

```

      K = J2+MIN(MAX(J1-J2-2,1),NS/2-3)
      DO 120 J = IU,NSRD
      IF (J.NE.IU) K = MOD(K+NS/2,NS)
      HS = YINT(ANG(K),05(K,2),4,4,PANG)
      HA = YINT(ANG(K),0A(K,2),4,4,PANG)
      HP = YINT(ANG(K),0P(K,2),4,4,PANG)
      TSTAR(I,J) = ((HP*(HA+HS)*ABEMF)/STFBOL)**0.25
      REMIND ITAPE
      RETURN
      END

```

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* THCF/CFFC43 \*\*\*\*\*

\*\*\*\*\* THCF/CFFC43 \*\*\*\*\*  
FOR S ME\*NASAS.THCF/CFFC43,ME\*NASAS.THCF/CFFC43  
FOR S9A-07/13/72-20:58:14 (0,)

FUNCTION THCF ENTRY POINT 000013

STORAGE USED: CODE(1) 000015; DATA(0) 000010; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000003 INJPS 0000 R 000000 THCF 0000 R 000001 X1 0000 R 000002 X2

00101	1*		FUNCTION THCF(RHO,T)
00101	2*	C	
00101	3*	C	THIS FUNCTION SUBPROGRAM COMPUTES :
00101	4*	C	THERMAL CONDUCTIVITY AS A FUNCTION OF DENSITY (SLUG/CU.FT)
00101	5*	C	AND TEMPERATURE (R) OF FC-43
00101	6*	C	UNITS BTU/(HR.FT.R)
00101	7*	C	
00103	8*		DATA X1,X2 /0.061442,-2.5E-05/
00106	9*		THCF = X1*X2*T
00107	10*		RETURN
00110	11*		END

END OF COMPILATION: NO DIAGNOSTICS.

DHD6\*P \*\*\*\*\* THCF/CFFC75 \*\*\*\*\*



FUNCTION THCF ENTRY POINT 000013

STORAGE USED: CODE(1) 000015; DATA(0) 000010; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000003 INJ\$ 0000 R 000000 THCF 0000 R 000001 X1 0000 R 000002 X2

```

00101 1*          FUNCTION THCF(RHO,T)
00101 2*          C
00101 3*          C THIS FUNCTION SUBPROGRAM COMPUTES :
00101 4*          C THERMAL CONDUCTIVITY AS A FUNCTION OF DENSITY (SLUG/CU.FT)
00101 5*          C AND TEMPERATURE (R) OF FC-75
00101 6*          C UNITS BTU/(HR.FT.R)
00101 7*          C
00103 8*          DATA X1,X2 /0.114181,-6.53E-05/
00106 9*          THCF = X1*X2*T
00107 10*         RETURN
00110 11*         END

```

END OF COMPILATION: NO DIAGNOSTICS.

QHD6,P \*\*\*\*\* THCF/CFHE \*\*\*\*\*

\*\*\*\*\* THCF/CFHE \*\*\*\*\*

QFOR'S ME\*NASAS5.THCF/CFHE,ME\*NASAS5.THCF/CFHE  
FOR S9A-07/13/72-20:58:18 (0,)

FUNCTION THCF ENTRY POINT 000022

STORAGE USED: CODE(1) 000024; DATA(0) 000015; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3S

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000006 DT 0000 000010 INJPS 0000 R 000000 THCF 0000 R 000007 THETA 0000 R 000005 TO  
0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3 0000 R 000004 X4

00101 1\* FUNCTION THCF(RHO,T)

00101 2\* C

00101 3\* C

00101 4\* C

00101 5\* C

00101 6\* C

00101 7\* C

00103 8\*

00103 9\*

00112 10\*

00113 11\*

00114 12\*

00115 13\*

THIS FUNCTION SUBPROGRAM COMPUTES :  
THERMAL CONDUCTIVITY AS A FUNCTION OF DENSITY (SLUG/CU.FT)  
AND TEMPERATURE (R) OF HELIUM  
UNITS BTU/(HR.FT.R)

DATA X1,X2,X3,X4 /0.0404,0.0302,-0.0033,0.0003/,T0,DT /160.0,20  
10.0/

THETA = (T-T0)/DT

THCF = ((X4\*THETA\*X3)\*THETA\*X2)\*THETA\*X1

RETURN

END

END OF COMPILATION: NO DIAGNOSTICS.

CHD6,P \*\*\*\*\* THCF/CFNAK \*\*\*\*\*

\*\*\*\*\* THCF/CFNAK \*\*\*\*\*

9FOR,S ME\*NASAS,THCF/CFNAK,ME\*NASAS,THCF/CFNAK  
FOR S9A-07/13/72-20:58:20 (0,)

FUNCTION THCF ENTRY POINT 000024

STORAGE USED: CODE(1) 000026; DATA(0) 000016; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000007 DT 0000 000011 INJPS 0000 R 000000 THCF 0000 R 000010 THETA 0000 R 000006 T0  
0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3 0000 R 000004 X4 0000 R 000005 X5

00101 1\* FUNCTION THCF(RHO,T)  
00101 2\* C  
00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
00101 4\* C THERMAL CONDUCTIVITY AS A FUNCTION OF DENSITY (SLUG/CU.FT)  
00101 5\* C AND TEMPERATURE (R) OF NAK 78.6  
00101 6\* C UNITS BTU/(HR.FT.R)  
00101 7\* C  
00101 8\* DATA X1,X2,X3,X4,X5 /13.36,1.414167,-0.142083,-0.069167,0.007083/  
00103 9\* 170,DT /659.67,300.0/  
00113 10\* THETA = (T-T0)/DT  
00114 11\* THCF = (((X5\*THETA\*X4)\*THETA\*X3)\*THETA\*X2)\*THETA\*X1  
00115 12\* RETURN  
00116 13\* END

END OF COMPILATION: NO DIAGNOSTICS.

9HDG,P \*\*\*\*\* THCF/CFSIL \*\*\*\*\*

FOR S ME\*NASAS5.THCF/CFSIL,ME\*NASAS5.THCF/CFSIL  
 FOR S9A-07/13/72-20:58:22 (0,)

FUNCTION THCF ENTRY POINT 000043

STORAGE USED: CODE(1) 000047; DATA(0) 000031; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDUS  
 0004 NIO2S  
 0005 NERR3S

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000003 1F 0000 000023 INJPS 0000 R 000000 THCF 0000 R 000001 X1 0000 R 000002 X2

00101 1\* FUNCTION THCF(RHO,T)

00101 2\* C  
 00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
 00101 4\* C THERMAL CONDUCTIVITY AS A FUNCTION OF DENSITY (SLUG/CU.FT) AND  
 00101 5\* C TEMPERATURE (R) OF DOW CORNING 200 SILICON OIL (1 CS)  
 00101 6\* C UNITS BTU/(HR.FT.R)  
 00101 7\* C TEMPERATURE .GE. 359.67 AND .LE. 859.67  
 00101 8\* C  
 00101 9\* C DATA X1,X2 /0.070052,-2.2105E-05/  
 00106 10\* THCF = X1\*X2\*T  
 00107 11\* IF (T.GE.359.67.AND.T.LE.859.67) RETURN  
 00111 12\* WRITE (6,1) T  
 00114 13\* 1 FORMAT (1H0,61HTHERMAL CONDUCTIVITY OF SILICON OIL, TEMP. OUT OF R  
 00114 14\* RANGE, T = ,F8.3,/)   
 00115 15\* RETURN  
 00116 16\* END

END OF COMPILATION: NO DIAGNOSTICS.

BHDG:P \*\*\*\*\* THCFN/FNAL \*\*\*\*\*

FOR S ME\*NASAS.THCFN/FNAL,ME\*NASAS.THCFN/FNAL  
 FOR S9A-07/13/72-20:58:25 (0,)

FUNCTION THCFN ENTRY POINT 000054

STORAGE USED: CODE(1) 000060; DATA(0) 000037; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)  
 0003 NWDJ5  
 0004 NI025  
 0005 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)  
 0000 000011 IF 0000 R 000007 DT 0000 000031 INJPS 0000 R 000000 THCFN 0000 R 000010 THETA  
 0000 R 000006 T0 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3 0000 R 000004 X4  
 0000 R 000005 X5

00101 1\* FUNCTION THCFN(T)  
 00101 2\* C  
 00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
 00101 4\* C THERMAL CONDUCTIVITY AS A FUNCTION OF TEMPERATURE (R) OF  
 00101 5\* C ALUMINUM 7075  
 00101 6\* C UNITS BTU/(HR.FT.R)  
 00101 7\* C TEMP. GE. 300.0.AND.LE. 1200.0 R  
 00101 8\* C  
 00101 9\* C DATA X1,X2,X3,X4,X5 /88.5,13.0665,0.33275,-1.758,0.25375/,T0,DT /  
 00103 10\* C 1400.0,200.0/  
 00113 11\* C THETA = (T-T0)/DT  
 00114 12\* C THCFN = (((X5\*THETA\*X4)\*THETA\*X3)\*THETA\*X2)\*THETA\*X1  
 00115 13\* C IF (T.GE.300.0.AND.T.LE.1200.0) RETURN  
 00117 14\* C WRITE (6,1) T  
 00122 15\* C 1 FORMAT (1H0,5HTHERMAL CONDUCTIVITY OF ALUMINUM, TEMP. OUT OF RANG  
 00122 16\* C IE, T = ,F8.3,//  
 00123 17\* C RETURN  
 00124 18\* C END

END OF COMPILATION: NO DIAGNOSTICS.

ENDG:P \*\*\*\*\* THCFN/FNBR \*\*\*\*\*

\*\*\*\*\* THCFN/FNBR \*\*\*\*\*

OFOR'S ME\*NASAS,THCFN/FNBR,ME\*NASAS5,THCFN/FNBR  
FOR S9A-07/13/72-20:58:27 (0,)

FUNCTION THCFN ENTRY POINT 000054

STORAGE USED: CODE(1) 000060; DATA(0) 000037; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDUS  
0004 NIO2S  
0005 NERR3S

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000011 IF 0000 R 000007 OT 0000 000031 INJPS 0000 R 000000 THCFN 0000 R 000010 THETA  
0000 R 000006 TO 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3 0000 R 000004 X4  
0000 R 000005 X5

FUNCTION THCFN(T)

```

00101 1* C
00101 2* C
00101 3* C THIS FUNCTION SUBPROGRAM COMPUTES :
00101 4* C THERMAL CONDUCTIVITY AS A FUNCTION OF TEMPERATURE (R) OF
00101 5* C BERYLIUM WITH 0.84-1.68 DEO
00101 6* C UNITS BTU/(HR.FT.R)
00101 7* C TEMP. 6E. 400.0 AND LE. 1700.0 R
00101 8* C
00101 9* C DATA X1,X2,X3,X4,X5 /108.863,-10.5643,0.82683,-0.20167,0.020167/,
00103 10* C 1T0,DT /400.0,200.0/
00113 11* C THETA = (T-T0)/DT
00114 12* C THCFN = ((X5*THETA+X4)*THETA+X3)*THETA+X2)*THETA+X1
00115 13* C IF (T.6E.400.0.AND.T.LE.1700.0) RETURN
00117 14* C WRITE (6,1) T
00122 15* C 1 FORMAT (1H0,58HTHERMAL CONDUCTIVITY OF BERYLIUM, TEMP. OUT OF RANG
00122 16* C 1E, T = ,F8.3,/)
00123 17* C RETURN
00124 18* C END
    
```

END OF COMPILATION: NO DIAGNOSTICS.

QHD6,P \*\*\*\*\* THCFN/FNCU \*\*\*\*\*



FOR,S ME\*NASAS5,THCMP/MPAL,ME\*NASAS5,THCMP/MPAL  
 FOR S9A-07/13/72-20:58:31 (0,)

FUNCTION THCMP ENTRY POINT 000054

STORAGE USED: CODE(1) 000060; DATA(0) 000037; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDJ\$  
 0004 N102\$  
 0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000011 IF 0000 R 000007 DT 0000 000031 INJPS 0000 R 000000 THCMP 0000 R 000010 THETA  
 0000 R 000006 T0 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3 0000 R 000004 X4  
 0000 R 000005 X5 0000 R 000005 X5

```

00101 1* FUNCTION THCMP(T)
00101 2* C
00101 3* C THIS FUNCTION SUBPROGRAM COMPUTES :
00101 4* C THERMAL CONDUCTIVITY AS A FUNCTION OF TEMPERATURE (R) OF
00101 5* C ALUMINUM 7075
00101 6* C UNITS BTU/(HR.FT.R)
00101 7* C TEMP. GE. 300.0.AND.LE. 1200.0 R
00101 8* C
00103 9* DATA X1,X2,X3,X4,X5 /88.5,13.0665,0.33275,-1.758,0.25375/,T0,DT /
00103 10* 1400.0,200.0/
00113 11* THETA = (T-T0)/DT
00114 12* THCMP = ((X5*THETA+X4)*THETA+X3)*THETA+X2)*THETA+X1
00115 13* IF (T.GE.300.0.AND.T.LE.1200.0) RETURN
00117 14* WRITE (6,1) T
00122 15* 1 FORMAT (1H0,58HTHERMAL CONDUCTIVITY OF ALUMINUM, TEMP. OUT OF RANG
00122 16* 1E, T = ,F8.3,/)
00123 17* RETURN
00124 18* END
  
```

END OF COMPILATION: NO DIAGNOSTICS.

END \*\*\*\*\* THCMP/MPBR \*\*\*\*\*



QFOR>S ME\*NASAS.THCMP/MPBR;ME\*NASAS.THCMP/MPBR  
 FOR S9A-07/13/72-20:58:34 (0\*)

FUNCTION THCMP ENTRY POINT 000054

STORAGE USED: CODE(1) 000060; DATA(0) 000037; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDJ\$  
 0004 NI02\$  
 0005 MERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000011 1F 0000 R 000007 DT 0000 000031 INJPS 0000 R 000000 THCMP 0000 R 000010 THETA  
 0000 R 000006 T0 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3 0000 R 000004 X4  
 0000 R 000005 X5

00101 1\* FUNCTION THCMP(IT)  
 00101 2\* C  
 00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
 00101 4\* C THERMAL CONDUCTIVITY AS A FUNCTION OF TEMPERATURE (R) OF  
 00101 5\* C BERYLIUM WITH 0.84-1.68 BEO  
 00101 6\* C UNITS BTU/(HR.FT.R)  
 00101 7\* C TEMP. 6E. 400.0 AND LE. 1700.0 R  
 00101 8\* C  
 00101 9\* C  
 00103 10\* DATA X1,X2,X3,X4,X5 /108.863,-10.5643,0.82683,-0.20167,0.020167/,  
 00113 11\* THETA = (T-T0)/DT  
 00114 12\* THCMP = ((X5\*THETA+X4)\*THETA+X3)\*THETA+X2)\*THETA+X1  
 00115 13\* IF (T.6E.400.0.AND.T.LE.1700.0) RETURN  
 00117 14\* WRITE (6,1) T  
 00122 15\* 1 FORMAT (1H0,58HTHERMAL CONDUCTIVITY OF BERYLIUM, TEMP. OUT OF RANG  
 00122 16\* 1E, T = ,F8.3,/)   
 00123 17\* RETURN  
 00124 18\* END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG:P \*\*\*\*\* THCMP/MPCU \*\*\*\*\*

\*\*\*\*\* THCMP/MPCU \*\*\*\*\*

QFOR.S ME\*NASAS.THCM/MPCU\*ME\*NASAS.THCM/MPCU  
 FOR 59A-07/13/72-20:58:36 (0,)

FUNCTION THCMP ENTRY POINT 000051

STORAGE USED: CODE(1) 000055; DATA(0) 000034; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDJ\$  
 0004 NIO2\$  
 0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000007 IF 0000 R 000005 DT 0000 000026 INJPS 0000 R 000000 THCMP 0000 R 000006 THETA  
 0000 R 000004 T0 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3

```

00101 1*
00101 2*
00101 3*
00101 4*
00101 5*
00101 6*
00101 7*
00103 8*
00111 9*
00112 10*
00113 11*
00115 12*
00120 13*
00120 14*
00121 15*
00122 16*

FUNCTION THCMP(T)
C
C THIS FUNCTION SUBPROGRAM COMPUTES :
C THERMAL CONDUCTIVITY AS A FUNCTION OF TEMPERATURE (R) OF COPPER
C UNITS BTU/HR.FT.R
C TEMP. GE. 500.0 AND LE. 1800.0 R
C
DATA X1,X2,X3 /228.369,-2.62067,-0.04033/,T0,DT /600.0,200.0/
THETA = (T-T0)/DT
THCMP = (X3*THETA*THETA+X2)*THETA+X1
IF (T.GE.500.0.AND.T.LE.1800.0) RETURN
WRITE (6,1) T
1 FORMAT (I10.56)THERMAL CONDUCTIVITY OF COPPER, TEMP. OUT OF RANGE,
1 T = ,F8.3//
RETURN
END
  
```

END OF COMPILATION: NO DIAGNOSTICS.

QMD6.P \*\*\*\*\* THCTB/TBAL \*\*\*\*\*

\*\*\*\*\* THCTB/TBAL \*\*\*\*\*

QFOR,S ME\*NASAS5.THCTB/TBAL,ME\*NASAS5.THCTB/TBAL  
FOR S9A-07/13/72-20:58:40 (0.)

FUNCTION THCTB ENTRY POINT 000054

STORAGE USED: CODE(1) 000060; DATA(0) 000037; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDJ\$  
0004 NI02\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000011 1F 0000 R 000007 DT 0000 000031 INJPS 0000 R 000000 THCTB 0000 R 000010 THETA  
0000 R 000006 T0 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3 0000 R 000004 X4  
0000 R 000005 X5

00101 1\* FUNCTION THCTB(T)  
00101 2\* C  
00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
00101 4\* C THERMAL CONDUCTIVITY AS A FUNCTION OF TEMPERATURE (R) OF  
00101 5\* C ALUMINUM 7075 UNITS BTU/(HR.FT.R)  
00101 6\* C  
00101 7\* C TEMP. GE. 300.0.AND.LE. 1200.0 R  
00101 8\* C  
00101 9\* C  
00103 10\* DATA X1,X2,X3,X4,X5 /88.5,13.0665,0.33275,-1.758,0.25375/,T0,DT /  
00103 11\* 1400.0,200.0/  
00113 12\* THETA = (T-T0)/DT  
00114 13\* THCTB = ((X5\*THETA+X4)\*THETA+X3)\*THETA+X2)\*THETA+X1  
00115 14\* IF (T.GE.300.0.AND.T.LE.1200.0) RETURN  
00117 15\* \*RITE (6,1) T  
00122 16\* 1 FORMAT (I40,58)THERMAL CONDUCTIVITY OF ALUMINUM, TEMP. OUT OF RANG  
00122 17\* 1E, T = ,FB.3,/  
00123 18\* RETURN  
00124 19\* END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* THCTB/TBRR \*\*\*\*\*

QFOR S ME\*NASAS,THCTB/TBBR,ME\*NASAS5,THCTB/TBBR  
FOR S9A-07/13/72-20:58:42 (0,)

FUNCTION THCTB ENTRY POINT 000054

STORAGE USED: COJE(1) 000060; DATA(0) 000037; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDUS  
0004 NIO2\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000011 1F 0000 R 000007 DT 0000 000031 INJPS 0000 R 000000 THCTB 0000 R 000010 THETA  
0000 R 000006 T0 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3 0000 R 000004 X4  
0000 R 000005 X5

00101 1\* FUNCTION THCTB(T)  
00101 2\* C  
00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES:  
00101 4\* C THERMAL CONDUCTIVITY AS A FUNCTION OF TEMPERATURE (R) OF  
00101 5\* C BERYLIUM WITH 0.84-1.68 BEC  
00101 6\* C UNITS BTU/(HR.FT.R)  
00101 7\* C TEMP. 6E. 400.0 AND LE. 1700.0 R  
00101 8\* C  
00101 9\* DATA X1,X2,X3,X4,X5 /108.863,-10.5643,0.82683,-0.20167,0.020167,  
00103 10\* 1T0,DT /400.0,200.0/  
00113 11\* THETA = (T-T0)/DT  
00114 12\* THCTB = ((X5\*THETA\*X4)\*THETA\*X3)\*THETA\*X2)\*THETA\*X1  
00115 13\* IF (T.GE.400.0.AND.T.LE.1700.0) RETURN  
00117 14\* WRITE (6,1) T  
00122 15\* 1 FORMAT (1H0,58HTHERMAL CONDUCTIVITY OF BERYLIUM, TEMP. OUT OF RANG  
00122 16\* 1E, T = ,F8.3,/  
00123 17\* RETURN  
00124 18\* END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* THCTB/TBCU \*\*\*\*\*

QFOR,S ME\*NASAS,THCTB/TBCU,ME\*NASAS,THCTB/TBCU  
 FOR S9A-07/13/72-20:58:44 (0,)

FUNCTION THCTB ENTRY POINT 000051

STORAGE USED: CODE(1) 000055; DATA(0) 000034; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDUS  
 0004 NIO2S  
 0005 NERR3S

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000007 IF 0000 R 000005 DT 0000 000026 INJPS 0000 R 000000 THCTB 0000 R 000006 THETA  
 0000 R 000004 T0 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3

```

00101 1*          FUNCTION THCTB(T)
00101 2*          C
00101 3*          C      THIS FUNCTION SUBPROGRAM COMPUTES :
00101 4*          C      THERMAL CONDUCTIVITY AS A FUNCTION OF TEMPERATURE (R) OF COPPER
00101 5*          C      UNITS BTU/(HR.FT.R)
00101 6*          C      TEMP. 6E. 500.0 AND LE. 1800.0 R
00101 7*          C
00103 8*          C      DATA X1,X2,X3 /228.369,-2.62067,-0.04033/.T0,DT /600.0,200.0/
00111 9*          C      THETA = (T-T0)/DT
00112 10*         C      THCTB = (X3*THETA*THETA*X2)*THETA*X1
00113 11*         C      IF (T.GE.500.0.AND.T.LE.1800.0) RETURN
00115 12*         C      WRITE (6,1) T
00120 13*         C      1 FORMAT (1H0,56HTHERMAL CONDUCTIVITY OF COPPER, TEMP. OUT OF RANGE,
00120 14*         C      1 T = ,F8.3,/)
00121 15*         C      RETURN
00122 16*         C      END
  
```

END OF COMPILATION: NO DIAGNOSTICS.

QHDG:P \*\*\*\*\* TK \*\*\*\*\*

QFOR,S ME\*NASAS.TK,ME\*NASAS.TK  
 FOR S9A-07/13/72-20:58:46 (0.)

FUNCTION TK ENTRY POINT 000211

STORAGE USED: CODE(1) 000245; DATA(0) 000062; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 ELAS  
 0004 ALOG  
 0005 NEXP65  
 0006 SORT  
 0007 NWDUS  
 0010 NIO2\$  
 0011 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000112	1JL	0001	000127	14L	0001	000174	15L	0000	000020	16F	0001	000201	17L
0000	R	000003	B	0000	R	000001	C	0000	R	000017	CC	0000	R	000015
0000	R	000002	EE	0003	R	000000	ELAS	0000	R	000005	F	0000	R	000040
0000	I	000012	K3	0000	R	000007	P	0000	R	000010	Q	0000	R	000011
0000	R	000000	TK	0000	R	000013	TN	0000	R	000014	V	0000	R	000015

00101	1*	FUNCTION	TK(GAMMA,A,BETA,DENSM,THETA,PHI,AN,ALPHA,VELM,
00101	2*	1	PO,TAU,DENST,W,TNN,AMAN,TIN,ROUT)
00101	3*	C	
00101	4*	C	THIS FUNCTION SUBPROGRAM COMPUTES :
00101	5*	C	THE THICKNESS OF THE PROTECTION LAYER TK , IN INCHES
00101	6*	C	
00103	7*		DATA C/1.2407013/
00105	8*		EE = 1./((3.*BETA)
00106	9*		B = GAMMA*A
00107	10*		D = 0.0436332*TNN*W
00110	11*		F = (ALPHA*TAU/(-ALOG(PO)))*EE
00111	12*		G = (1./DENSM)**.33333
00112	13*		P = (DENSM/DENST)**PHI
00113	14*		Q = (VELW/(12.0*SORT(ELAS(TIN)*32.174/DENST)))*THETA
00114	15*		S = (2./((3.*AN*THETA*BETA+2.))*EE
00115	16*		KG = 0
00116	17*		TN = 0.*5
00117	18*		13 KG = KG+1
00120	19*		IF (KG.EQ.1) GO TO 14
00122	20*		IF (KG.GT.10) GO TO 15
00124	21*		TN = TK
00125	22*		14 V = (D*(ROUT+TN)+AMAN)
00126	23*		E = V**EE
00127	24*		R = B*C*E*F*G*P*Q*S
00130	25*		TK = TN-((TN-R)/(1.-(EE*R*D/V)))

\*\*\*\*\* TK \*\*\*\*\*

00131  
00132  
00133  
00134  
00135  
00137  
00140  
00141

26\*  
27\*  
28\*  
29\*  
30\*  
31\*  
32\*

CC = TK/TN-1.  
IF (ABS(CC).LT..0001) GO TO 17  
GO TO 13  
15 WRITE (6,16)  
16 FORMAT (29HEGN FOR THICK DOESNT CONVERGE)  
17 RETURN  
END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG:P \*\*\*\*\* TNH \*\*\*\*\*

QFOR,S ME\*NASAS.TNH,ME\*NASAS.TNH  
FOR S9A-07/13/72-20:58:48 (0,)

FUNCTION TNH ENTRY POINT 000056

STORAGE USED: CODE(1) 000064; DATA(0) 000030; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 CPAIR  
0004 ENTAIR  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000037 10L 0001 000004 106G 0001 000041 30L 0003 R 000000 CPAIR 0000 R 000014 DFTN  
0004 R 000000 ENTAIR 0000 R 000015 FTN 0000 000020 INJPS 0000 I 000013 N 0000 R 000001 TN  
0000 R 000000 TNH

00101 1\* FUNCTION TNH(ENT)

00101 2\* C  
00101 3\* C  
00101 4\* C  
00101 5\* C  
00101 6\* C

THIS FUNCTION SUBPROGRAM COMPUTES :  
TEMPERATURE OF THE ATMOSPHERE AS A FUNCTION OF ENTHALPY (BTU/LBM)  
UNITS R

DIMENSION TN(10)  
TN(1) = 2500.0  
DO 10 N = 1,10

DFTN = CPAIR(TN(N))  
FTN = ENTAIR(TN(N))-ENT  
TN(N+1) = TN(N)-FTN/DFTN  
IF (N.EQ. 1) GO TO 10  
IF (ABS(TN(N+1)-TN(N)) .LE. 2.0) GO TO 30  
10 CONTINUE  
30 TNH = TN(N+1)  
RETURN  
END

END OF COMPILATION: NO DIAGNOSTICS.

ENDG/P \*\*\*\*\* TRMATX \*\*\*\*\*



\*\*\*\*\* TRMATX \*\*\*\*\*

FOR S ME\*NASAS,TRMATX,ME\*NASAS,TRMATX  
FOR S9A-07/13/72-20:58:50 (0,)

SUBROUTINE TRMATX ENTRY POINT 000262

STORAGE USED: CODE(1) 000275; DATA(10) 000051; BLANK COMMON(2) 000000

COMMON BLOCKS:

0003 ORD 003721  
0004 ABSRST 000601

EXTERNAL REFERENCES (BLOCK, NAME)

0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000003	106G	0001	000004	111G	0001	000021	120G	0001	000022	123G	0001	000053	134G		
0001	000056	137G	0001	000066	142G	0001	000121	153G	0001	000137	162G	0001	000145	166G		
0001	000154	171G	0001	000211	201G	0001	000212	204G	0001	000217	211G	0001	000224	216G		
0004	R	000000	ALPHFN	0004	R	000226	ALPHMP	0003	R	003465	CSF	0000	R	003464	DXX2	
0003	003472	DXX21	0003	003463	DZMFN	0003	003410	EBFN	0003	003441	EBMP	0004	000461	EMITFN		
0004	000512	EMITMP	0003	003467	EXTIFN	0003	003471	EXTIMP	0003	003466	EXTSFN	0003	003470	EXTSMP		
0004	000517	EXTVCT	0000	I	000000	I	0000	00027	INJPS	0000	I	000001	J	000007	JK	
0000	I	000002	K	0000	I	000004	L	0003	003461	LCT	0003	003462	LTI	0003	003460	WCVRD
0000	I	000003	M1	0000	I	000006	M2	0003	R	003473	SS	0003	R	003670	SSTT	TRMATX
0003	003446	XX2	0003	003453	ZZ2											

SUBROUTINE TRMATX

```

00101 1*
00101 2*
00101 3*
00101 4*
00101 5*
00101 6*
00101 7*
00103 8*
00103 9*
00103 10*
00104 11*
00104 12*
00105 13*
00110 14*
00113 15*
00115 16*
00117 17*
00122 18*
00125 19*
00127 20*
00130 21*

C
C
C
C
C
C
COMMON /ORD/ TRMTX(30,60),EBFN(5,5),EBMP(5),XX2(5),ZZ2(5),
1 MCVRD,LCT,LTI,DZMFN,
2 DXX2,CSF,EXTSFN,EXTIFN,EXTSMP,EXTIMP,DXX21,SS(5,5,5)
3 ,SSTT(5,5)
COMMON /ABSRST/ALPHFN(5,5,6),ALPHMP(5,31),EMITFN(5,5),EMITMP(5),
1 EXTVCT(50)
DO 2 I=1,25
DO 1 J=1,25
1 TRMTX(I,J)=0.0
2 TRMTX(I,I)=1.0
DO 4 I=26,30
DO 3 J=26,30
3 TRMTX(I,J)=-(1.0-ALPHMP(I-25,J))*SSTT(I-25,J-25)*0.25
TRMTX(I,26)=TRMTX(I,26)/2.0
TRMTX(I,30)=TRMTX(I,30)/2.0

```

```

***** TRMATX *****
00131 22* 4 TRMTX(I,I)=1.0+TRMTX(I,I)
00133 23* 00 5 I=1,5
00136 24* 00 5 J=1,5
00141 25* 00 5 K=1,5
00144 26* M1 =25*K
00145 27* L =(I-1)*5 +J
00146 28* 5 TRMTX(L,M1) =-(1.0-ALPHFN(I,J,K))*SS(I,J,K)/4.0
00152 29* 00 6 I=1,25
00155 30* TRMTX(I,26) =TRMTX(I,26)/2.0
00156 31* 6 TRMTX(I,30) =TRMTX(I,30)/2.0
00160 32* CSFZX = CSF*DX2/4.0
00161 33* 00 7 I=1,5
00164 34* M2 =25*I
00165 35* 00 7 J=1,5
00170 36* 00 7 K=1,5
00173 37* JK =(J-1)*5+K
00174 38* 7 TRMTX(M2,JK) =-(1.0-ALPHMP(I,JK))*SS(J,K,I)*CSFZX
00200 39* 00 10 I=26,30
00203 40* 00 8 K=21,25
00206 41* 8 TRMTX(I,K) =TRMTX(I,K)/2.0
00210 42* 00 9 J=1,21,5
00213 43* 9 TRMTX(I,J) =TRMTX(I,J)/2.0
00215 44* 00 10 J=5,25,5
00220 45* 10 TRMTX(I,J) =TRMTX(I,J)/2.0
00223 46* RETURN
00224 47* END

```

END OF COMPILATION: NO DIAGNOSTICS.

END6,P \*\*\*\*\* TRNSPT \*\*\*\*\*

OFOR,S ME\*NASAS5.TRNSPT,ME\*NASAS5.TRNSPT  
FOR S9A-07/13/72-20:58:53 (0,)

SUBROUTINE TRNSPT ENTRY POINT 000143

STORAGE USED: CODE(1) 000171; DATA(0) 000040; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NEXP6\$  
0004 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000062 10L 0001 000106 15L 0001 000050 5L 0000 000030 INJPS 0000 R 000000 NUL  
0000 R 000001 XX

```

00101 1* SUBROUTINE TRNSPT (RE,PRL,DELTA,FR,FNU) 01
00101 2* C
00101 3* C THIS SUBROUTINE COMPUTES :
00101 4* C THE NON-DIMENSIONAL FRICTION FACTOR FR ,AND THE NON-DIMENSIONAL
00101 5* C CONVECTIVE FILM COEFFICIENT FNU
00101 6* C
00101 7* C REAL NUL
00103 8* C
00104 9* IF (RE.LT.2300.0) GO TO 10
00106 10* NUL = 0.116*(RE**0.667-125.0)*PRL**0.3333*(1.0*DELTA**0.667)
00107 11* IF (RE.GT.1.0E*06) GO TO 5
00111 12* FR = 10.0054+0.396/RE**0.3)/DELTA
00112 13* GO TO 15
00113 14* 5 FR = (0.0032+0.222/RE**0.237)/DELTA
00114 15* GO TO 15
00115 16* 10 FR = 64.0/(RE*DELTA)
00116 17* XX = RE*PRL*DELTA
00117 18* NUL = 3.65+0.0668*XX/(1.0+0.045*XX**0.667)
00120 19* 15 IF (PRL.LT.0.1) NUL = 6.5+0.025*(RE*PRL)**0.8
00122 20* FNU = 4.0*NUL/(DELTA*RE*PRL)
00123 21* RETURN
00124 22* END

```

END OF COMPILATION: NO DIAGNOSTICS.

ENDG,P \*\*\*\*\* TTIP \*\*\*\*\*

\*\*\*\*\* TTIP \*\*\*\*\*

QFOR,S ME\*NASAS.TTIP,ME\*NASAS.TTIP  
FOR S9A-07/13/72-20:58:55 (0,)

SUBROUTINE TTIP ENTRY POINT 000043

STORAGE USED: CODE(1) 000060; DATA(0) 000101; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 POLY  
0004 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001 000005 1166 0001 000007 1216 0000 R 000043 AA 0000 R 000000 CC  
0000 I 000057 I 0000 000067 INJPS 0000 I 000060 J 0003 R 000000 POLY 0000 R 000052 COEF

00101 1\* SUBROUTINE TTIP (NC,TSOTB,TLOTB)  
00103 2\* DIMENSION CC(5,7),AA(7),COEF(5)  
00104 3\* REAL NC  
00105 4\* DATA(CC(1,2),I=1,5)/-.4815,-.8920814,4.894438,-14.89583,10.39930/,  
00105 5\* (CC(1,3),I=1,5)/.53650067,7351966,-12.35190,40.88471,-30.657R2/,  
00105 6\* (CC(1,4),I=1,5)/-.440667,2304286,11.64157,-44.56229,35.2602R/,  
00105 7\* (CC(1,5),I=1,5)/.212667,-.4978796,-5.648548,24.37024,-19.93047/,  
00105 8\* (CC(1,6),I=1,5)/-.05333347,.1983361,40554,-6.58330,5.48609/,  
00105 9\* (CC(1,7),I=1,5)/.00533335,-.02525962,-.139998,.6925878,-.5833303/  
00114 10\* AA(1) = 1.0  
00115 11\* DO 20 I=2,7  
00120 12\* DO 10 J=1,5  
00123 13\* 10 COEF(J) = CC(J,I)  
00125 14\* 20 AA(I) = POLY (5,COEF,TSOTB)  
00127 15\* TLOTB = POLY (7,AA,NC)  
00130 16\* RETURN  
00131 17\* END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* TTIPS \*\*\*\*\*

DFOR,S ME\*NASAS.TTIPS,ME\*NASAS.TTIPS  
FOR S9A-07/13/72-20:58:57 (0,)

SUBROUTINE TTIPS ENTRY POINT 000345

STORAGE USED: CODE(1) 000366; DATA(0) 000110; BLANK COMMON(2) 000000

COMMON BLOCKS:

0003 AOBH 000352  
0004 GIN 001610

EXTERNAL REFERENCES (BLOCK, NAME)

0005 NUSA  
0006 EFFICY  
0007 TTIP  
0010 EXP  
0011 NEXP6\$  
0012 NPRT\$  
0013 NI02\$  
0014 NSTOP\$  
0015 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000041 105F 0001 000303 110L 0001 000136 136G  
0000 R 000033 AA 0003 R 000001 AL 0000 R 000021 BB1  
0003 R 000161 CVFL 0003 R 000173 CONFNA 0003 R 000147 CP  
0000 R 000030 C4 0000 R 000020 C5 0003 R 000000 D  
0000 R 000036 DTB 0003 R 000326 DTL 0003 R 000244 EFF  
0000 R 000027 ETAPR2 0000 R 000024 ET1A 0000 R 000026 ETA2  
0003 000231 HAB 0000 000064 INJP\$ 0000 I 000017 L  
0000 R 000003 NCPR2 0000 R 000000 NC1 0000 R 000001 NC2  
0004 001504 NTM 0005 R 000000 NUSA 0000 R 000004 PI  
0004 000454 QIFN 0004 001274 GITB 0003 000314 GOUT  
0004 001506 QIO 0000 R 000010 REY 0000 R 000005 SIGMA  
0000 R 000032 TB4 0003 R 000026 TH 0003 R 000040 TIN  
0000 R 000040 TLOTB2 0004 000000 TM 0003 000340 TOUT  
0000 R 000016 TSTR 0000 R 000015 TS4 0004 001607 TX  
  
0001 000126 30L 0001 000120 20L  
0000 R 000006 COEF 0000 R 000022 BB2  
0000 R 000014 C2 0000 R 000015 C1  
0000 R 000011 DOL 0000 R 000035 DFNTB  
0000 R 000025 ETAPR1 0003 R 000205 EWIS  
0003 000013 H 0000 R 000034 FNTB  
0000 R 000002 NCPR1 0003 I 001605 NSRD  
0003 000076 NT 0003 000052 PIN  
0000 R 000007 PR 0004 000764 OSTB  
0000 R 000031 TB3 0000 R 000217 TB  
0000 R 000037 TLOTB1 0003 R 000270 TL  
0003 R 000077 TSTAR 0000 R 000023 TSOTB  
0003 R 000135 VSC 0000 R 000012 U

00101 1\* SUBROUTINE TTIPS(IT,ITST,H1,H2,TL1,TL2)  
00103 2\* PARAMETER NTP = 10  
00104 3\* PARAMETER NTP1 = NTP+1  
00105 4\* REAL M  
00106 5\* COMMON /ADBH/ D,AL(NTP),H(NTP1),H(NTP),TH(NTP),TIN(NTP),PIN(NTP),  
00106 6\* M(NTP),NT,ITSTAR(NTP,3),VSC(NTP),CP(NTP),CONFL(NTP),  
00106 7\* CONFNA(NTP),EMIS(NTP),TB(NTP),HAB(NTP1),EFF(NTP,2),TL(NTP,2),  
00106 8\* GOUT(NTP),DTL(NTP),TOUT(NTP)  
00107 9\* COMMON /GIN/ TM(100),QSFN(100,2),QIFN(100,2),OSTB(100,2),

```

***** TTIPS *****
00107 10*      1  Q1TB(100,2),NTM,NSRD,GT0,TX
00110 11*      REAL NUSA,NC1,NC2,NCPR1,NCPR2
00111 12*      DATA PI/3.1415927/
00113 13*      DATA SIGMA/1.714E-9/
00115 14*      COEF=EMIS(IT)*SIGMA/(CONFMA(IT)*TM(IT))
00116 15*      PR=VSC(IT)*CP(IT)/CONFL(IT)
00117 16*      REY=48*M(IT)/(PI*D*VSC(IT))
00120 17*      DOL=D/(12*AL(IT))
00121 18*      U=PI*CONFL(IT)*AL(IT)*NUSA(REY,PR,DOL)/(M(IT)*CP(IT))
00122 19*      C1=M(IT)*CP(IT)*(1-EXP(-U))
00123 20*      C2=NSRD*EMIS(IT)*SIGMA*AL(IT)/12
00124 21*      TB(IT)=.9*TM(IT)
00125 22*      IF(NSRD.NE.2)GO TO 20
00127 23*      TS4=(TSTAR(IT,1)**4+TSTAR(IT,2)**4)/2
00130 24*      TSTR=TS4**25
00131 25*      GO TO 30
00132 26*      TS4=TSIAR(IT,IIST)**4
00133 27*      TSTR=TSTAR(IT,IIST)
00134 28*      CONTINUE
00135 29*      DO 100 L=2,10
00140 30*      C5=TB(IT)**2*COEF/12
00141 31*      B81=C5*H1**2
00142 32*      B82=C5*H2**2
00143 33*      NC1=TB(IT)*B81
00144 34*      NC2=TB(IT)*B82
00145 35*      NCPR1=3*B81
00146 36*      NCPR2=3*B82
00147 37*      TSOTB=TSTR/TB(IT)
00150 38*      CALL EFFICY(NC1,NCPR1,TSOTB,TB(IT),ETAL,ETAPR1)
00151 39*      EFF(IT,1) = ETAL*100.0
00152 40*      CALL EFFICY(NC2,NCPR2,TSOTB,TB(IT),ETA2,ETAPR2)
00153 41*      EFF(IT,2) = ETA2*100.0
00154 42*      C4=ETAL*H1+ETA2*H2
00155 43*      TB3=TB(IT)*3
00156 44*      TB4=TB3*TB(IT)
00157 45*      AA=TB4-TS4
00160 46*      FNTB=C1*(TM(IT)-TB(IT))-C2*C4*AA
00161 47*      DFNTB=-C1-C2*(4*TB3+C4*AA*(ETAPR1*H1+ETAPR2*H2))
00162 48*      DTB=FNTB/DFNTB
00163 49*      TB(IT)=TB(IT)-DTB
00164 50*      IF(ABS(DTB).LE.0.005)GO TO 110
00166 51*      CONTINUE
00170 52*      PRINT 105
00172 53*      FORMAT(' BASE TEMPERATURE DOES NOT CONVERGE')
00173 54*      STOP
00174 55*      CONTINUE
00175 56*      CALL TTIP(NC1,TSOTB,TLOTB1)
00176 57*      CALL TTIP(NC2,TSOTB,TLOTB2)
00177 58*      TL1=TLOTB1*(TB(IT)-TSTR)+TSTR
00200 59*      TL2=TLOTB2*(TB(IT)-TSTR)+TSTR
00201 60*      RETURN
00202 61*      END

```

END OF COMPILATION: NO DIAGNOSTICS.

\*\*\*\*\* TTIPS \*\*\*\*\*

QHDG/P \*\*\*\*\* VISC/CFFC43 \*\*\*\*\*

DATE 071372

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QFOR.S ME\*NASAS.VISC/CFFC43,ME\*NASAS.VISC/CFFC43  
 FOR S9A-07/13/72-20:59:00 (0.)

FUNCTION VISC ENTRY POINT 000032

STORAGE USED: CODE(1) 000036; DATA(0) 000021; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 EXP  
 0004 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000006 DT 0000 000012 INJPS 0000 R 000007 THETA 0000 R 000005 T0 0000 R 000000 VISC  
 0000 R 000010 VNU 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3 0000 R 000004 X4

00101 1\* FUNCTION VISC(RHO,T)

00101 2\* C  
 00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
 00101 4\* C VISCOSITY AS A FUNCTION OF DENSITY (SLUG/CU.FT)  
 00101 5\* C AND TEMPERATURE (R) OF FC-43  
 00101 6\* C UNITS SLUG/(FT.HR)  
 00101 7\* C

00103 8\* DATA X1,X2,X3,X4 /2.76,-2.043483,0.362,-0.034717/.T0,DT /439.67,90  
 00103 9\* 1.0/  
 00112 10\* THETA = (T-T0)/DT  
 00113 11\* VNU = EXP(((X4\*THETA+X3)\*THETA+X2)\*THETA+X1)  
 00114 12\* VISC = VNU\*RHO\*0.03875  
 00115 13\* RETURN  
 00116 14\* END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG/P \*\*\*\*\* VISC/CFFC75 \*\*\*\*\*



\*\*\*\*\* VISC/CF75 \*\*\*\*\*  
 QFOR,S ME\*NASAS.VISC/CF75,ME\*NASAS.VISC/CF75  
 FOR S9A-07/13/72-20:59:03 (0,)

FUNCTION VISC ENTRY POINT 000032

STORAGE USED: CODE(1) 000036; DATA(0) 000021; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 EXP  
 0004 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000006 DT 0000 000012 INJPS 0000 R 000007 THETA 0000 R 000005 T0 0000 R 000000 VISC  
 0000 R 000010 VNU 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3 0000 R 000004 X4

00101 1\* FUNCTION VISC(RHO,T)

00101 2\* C  
 00101 3\* C THIS FUNCTION SUBPROGRAM COMPUTES :  
 00101 4\* C VISCOSITY AS A FUNCTION OF DENSITY (SLUG/CU.FT)  
 00101 5\* C AND TEMPERATURE (R) OF FC-75  
 00101 6\* C UNITS SLUG/(FT.HR)  
 00101 7\* C

DATA X1,X2,X3,X4 /1.639,-1.312933,0.25265,-0.02471667/,T0,DT / 409

1.67,60.0/

THETA = (T-T0)/DT

VNU = EXP(((X4\*THETA+X3)\*THETA+X2)\*THETA+X1)

VISC = VNU\*RHO\*0.03875

RETURN

END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG,P \*\*\*\*\* VISC/CFHE \*\*\*\*\*

\*\*\*\*\* VISC/CFHE \*\*\*\*\*

CFOR,S ME\*NASAS.VISC/CFHE,ME\*NASAS.VISC/CFHE  
FOR S9A-07/13/72-20:59:04 (07)

FUNCTION VISC ENTRY POINT 000016  
STORAGE USED: CODE(1) 000022; DATA(0) 000011; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NEXP6\$  
0004 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000003 INJPS 0000 R 000000 VISC

00101	1*	C	FUNCTION VISC(RHO,T)
00101	2*	C	
00101	3*	C	THIS FUNCTION SUBPROGRAM COMPUTES :
00101	4*	C	VISCOSITY AS A FUNCTION OF DENSITY (SLUG/CU.FT)
00101	5*	C	AND TEMPERATURE (R) OF HELIUM
00101	6*	C	UNITS SLUG/(FT.HR)
00101	7*	C	
00103	8*		VISC = 2.58394E-05*T**0.647
00104	9*		RETURN
00105	10*		END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG.P \*\*\*\*\* VISC/CFNAK \*\*\*\*\*

\*\*\*\*\* VISC/CFNAK \*\*\*\*\*  
QFOR'S ME\*NASAS5.VISC/CFNAK\*ME\*NASAS5.VISC/CFNAK  
FOR S9A-07/13/72-20:59:07 (0,)

FUNCTION VISC ENTRY POINT 000025

STORAGE USED: CODE(1) 000027; DATA(0) 000017; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 R 000007 DT 0000 000012 INJPS 0000 R 000010 THETA 0000 R 000006 T0 0000 R 000000 VISC  
0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3 0000 R 000004 X4 0000 R 000005 X5

00101 1\* FUNCTION VISC(RHO,T)

00101 2\* C

00101 3\* C

00101 4\* C

00101 5\* C

00101 6\* C

00101 7\* C

00103 8\*

00103 9\*

00113 10\*

00114 11\*

00115 12\*

00116 13\*

THIS FUNCTION SUBPROGRAM COMPUTES :  
VISCOSITY AS A FUNCTION OF DENSITY (SLUG/CU.FT)  
AND TEMPERATURE (R) OF NAK 78.6  
UNITS SLUG/(FT.HR)

DATA X1\*X2\*X3\*X4\*X5 /1.316\*-0.896667\*0.419833\*-0.102833\*0.009667/.  
110.0T /659.67\*300.0/  
THETA = (T-10)/DT  
VISC = (((X5\*THETA+X4)\*THETA+X3)\*THETA+X2)\*THETA+X1)/32.174  
RETURN  
END

END OF COMPILATION: NO DIAGNOSTICS.

QHDG.P \*\*\*\*\* VISC/CFSIL \*\*\*\*\*

FOR,S ME\*NASAS,VISC/CFSIL,ME\*NASAS,VISC/CFSIL  
 FOR S9A-07/13/72-20:59:09 (0,)

FUNCTION VISC ENTRY POINT 000061

STORAGE USED: CODE(1) 000067; DATA(0) 000040; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 EXP  
 0004 NWDJ5  
 0005 NI025  
 0006 NERR35

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000 000011 IF 0000 R 000006 DT 0000 000030 INJPS 0000 R 000007 THETA 0000 R 000005 T0  
 0000 R 000000 VISC 0000 R 000010 VNU 0000 R 000001 X1 0000 R 000002 X2 0000 R 000003 X3  
 0000 R 000004 X4

```

00100 1* C
00100 2* C THIS FUNCTION SUBPROGRAM COMPUTES :
00101 3* C FUNCTION VISC(RHO,T)
00101 4* C VISCOSITY AS A FUNCTION OF DENSITY (SLUG/CU.FT) AND
00101 5* C TEMPERATURE (R) OF DOW CORNING 200 SILICON OIL (1 CS)
00101 6* C UNITS SLUG/(FT.HR)
00101 7* C TEMPERATURE .6E, 359.67 AND .LE, 859.67
00101 8* C
00103 9* C DATA X1,X2,X3,X4 /0.683,-1.0045,0.3065,-0.04/,T0,DT /459.67,100.0/
00112 10* C THETA = (T-T0)/DT
00113 11* C VNU = EXP(((X4*THETA+X3)*THETA+X2)*THETA+X1)
00114 12* C VISC = VNU*RH0*0.03875
00115 13* C IF (T.GE.360.67.AND.T.LE.860.67) RETURN
00117 14* C WRITE (6,1) T
00122 15* C 1 FORMAT (IHD,50HVISCOSITY OF SILICON OIL, TEMP. OUT OF RANGE, T = ,
00122 16* C 1F8.3,/)
00123 17* C RETURN
00124 18* C END
  
```

END OF COMPILATION: NO DIAGNOSTICS.

ENDG:P \*\*\*\*\* YINT \*\*\*\*\*

GFORS ME\*NASAS.YINT,ME\*NASAS.YINT  
FOR S9A-07/13/72-20:59:11 (0,)

FUNCTION YINT ENTRY POINT 000265

STORAGE USED: CODE(1) 000315; DATA(0) 000136; BLANK COMMON(2) 000000

EXTERNAL REFERENCES (BLOCK, NAME)

0003 NWDUS  
0004 NIO2\$  
0005 NERR3\$

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000075	1136	0001	000146	1306	0001	000172	1406	0001	000201	1436	0001	000215	1506	
0001	000106	2L	0001	000230	6L	0000	000060	7F	0001	000240	8L	0000	000072	9F	
0000	I	000053	I	0000	000111	INJPS	0000	I	000057	J	0000	I	000055	ME	
0000	I	000051	MM	0000	I	000054	M0	0000	I	000052	M1	0000	R	000025	X0
0000	R	000001	XX	0000	R	000000	YINT	0000	R	000037	YY	0000	R	000025	X0

```

00101 1* FUNCTION YINT(X,Y,N,M,P)
00101 2* C
00101 3* C THIS SUBROUTINE COMPUTES :
00101 4* C AITKEN INTERPOLATION
00101 5* C X ARRAY OF ABSCISSAE
00101 6* C Y ARRAY OF ORDINATES
00101 7* C N NUMBER OF ELEMENTS IN ARRAYS X AND Y
00101 8* C P X-VALUE OF INTERPOLATION
00101 9* C M DEGREE OF POLYNOMIAL PASSING THROUGH Y(X) IN THE VICINITY OF P
00101 10* C
00103 11* DIMENSION X(N),Y(N),XX(10),XP(10),XQ(10),YY(10)
00104 12* IF(P.LT.(2.*X(1)-X(2)).OR.P.GT.(2.*X(N)-X(N-1))) GO TO 8
00106 13* MM = MAX(MIN(M,10),2)
00107 14* IF (N.LT.MM.OR.N.LE.2) GO TO 6
00111 15* M1 = MM/2
00112 16* DO 1 I = 1,N
00115 17* IF (P .LE. X(I)) GO TO 2
00117 18* 1 CONTINUE
00121 19* 2 M0 = I-M1
00122 20* ME = M0+MM-1
00123 21* IF (M0 .LT. 1) M0 = 1
00125 22* IF (ME .GT. N) M0 = N-MM+1
00127 23* DO 3 I = 1,MM
00132 24* L = M0+I-1
00133 25* XP(I) = X(L)-P
00134 26* XQ(I) = X(L)
00135 27* 3 YY(I) = Y(L)
00137 28* DO 5 I = 2,MM
00142 29* DO 4 J = I,MM

```

```

***** YINT *****
00145 30* 4 XX(J) = (YY(I-1)*XP(J)-YY(J)*XP(I-1))/(XQ(J)-XQ(I-1))
00147 31* 00 5 J = 1,MM
00152 32* 5 YY(J) = XX(J)
00155 33* YINT = YY(NM)
00156 34* RETURN
00157 35* 6 *RITE (6,7)
00161 36* 7 FORMAT (/,1X,45HINTERPOL. IS IMPOSSIBLE, DATA ARRAY TOO SMALL)
00162 37* RETURN
00163 38* 8 *RITE (6,9) P
00166 39* 9 FORMAT (/,1X,29HINTERPOL. IS IMPOSSIBLE, P = ,E12.4,12HOUT OF RANG
00166 40* 1E)
00167 41* RETURN
00170 42* END

```

END OF COMPILATION: NO DIAGNOSTICS.

END OF COMPILATION: NO DIAGNOSTICS.

\*\*\*\*\* CFFC43 \*\*\*\*\*

DELT, L' ME\*NASAS, CFFC43  
ELT 005-07/13-20:59  
000001 000 CLASS  
000002 000 IN  
000003 000 DEF  
000004 000 END

CFFC43

BETA, CAPPA, CPF, HFL, RHOF, THCF, VISC  
BETA, CAPPA, CPF, HFL, RHOF, THCF, VISC

OHOG, P \*\*\*\*\* CFFC75 \*\*\*\*\*

\*\*\*\*\* CFFC75 \*\*\*\*\*

DELTA ME\*NASAS.CFFC75  
ELT 005-07/13-20:59  
000001 000 CLASS CFFC75  
000002 000 IN BETA,CAPPA,CPF,HFL,RHOF,THCF,VISC  
000003 000 DEF BETA,CAPPA,CPF,HFL,RHOF,THCF,VISC  
000004 000 END

OHG6/P \*\*\*\*\* CFHE \*\*\*\*\*



\*\*\*\*\* CFHE \*\*\*\*\*

QELT,L ME\*NASAS,CFHE  
 ELT 005-07/13-20:59  
 000001 000 CLASS CFHE  
 000002 000 IN BETA,CAPPA,CPF,HFL,PF,RHOF,THCF,VISC  
 000003 000 DEF BETA,CAPPA,CPF,HFL,PF,RHOF,THCF,VISC  
 000004 000 END

QHDG,P \*\*\*\*\* CFNAK \*\*\*\*\*

\*\*\*\*\* CFNAK \*\*\*\*\*

QELT>L ME\*NASAS.CFNAK  
ELY 005-07/13-20:59  
000001 000 CLASS  
000002 000 IN  
000003 000 DEF  
000004 000 END

CFNAK  
BETA,CAPPA,CPF,HFL,PF,RHOF,THCF,VISC  
BETA,CAPPA,CPF,HFL,PF,RHOF,THCF,VISC

QHDG,P \*\*\*\*\* CFSIL \*\*\*\*\*

\*\*\*\*\* CFSIL \*\*\*\*\*

GELT,L ME#NASAS,CFSIL  
 ELT 005-07/13-20:59  
 000001 000 CLASS  
 000002 000 IN  
 000003 000 DEF  
 000004 000 END

CFSIL

BETA,CAPPA,CPF,HFL,PF,RHOF,THCF,VISC  
 BETA,CAPPA,CPF,HFL,PF,RHOF,THCF,VISC

CHDS:P \*\*\*\*\* FNAL \*\*\*\*\*

\*\*\*\*\* FNAL \*\*\*\*\*

QELT,L ME=NASAS,FNAL  
ELT 005-07/13-20:59  
000001 000 CLASS FNAL  
000002 000 IN CPFN,DTHCFN,THCFN  
000003 000 DEF CPFN,DTHCFN,THCFN  
000004 000 END

QHDG,P \*\*\*\*\* FNBR \*\*\*\*\*

\*\*\*\*\* FNBR \*\*\*\*\*

QELT,L ME\*NASAS,FNBR  
 ELT 005-07/13-20:59  
 000001 000 CLASS FNBR  
 000002 000 IN CPFN,DTHCFN,THCFN  
 000003 000 DEF CPFN,DTHCFN,THCFN  
 000004 000 END

ENDG:P \*\*\*\*\* FNCU \*\*\*\*\*

\*\*\*\*\* FNCU \*\*\*\*\*

DELTL ME\*NASAS.FNCU  
ELT 005-07/13-20:59  
000001 000 CLASS FNCU  
000002 000 IN CPFN,DTMCFN,THCFN  
000003 000 DEF CPFN,DTMCFN,THCFN  
000004 000 END

\*\*\*\*\* MPAL \*\*\*\*\*

\*\*\*\*\* MPAL \*\*\*\*\*  
 QELT,L ME\*NASAS.MPAL  
 ELT 005-07/13-20:59  
 000001 000 CLASS  
 000002 000 IN  
 000003 000 DEF  
 000004 000 END

MPAL  
 CPMP,DTHCMP,ELAS,THCMP  
 CPMP,DTHCMP,ELAS,THCMP

ENDG,P \*\*\*\*\* MPBR \*\*\*\*\*

\*\*\*\*\* MPBR \*\*\*\*\*

DELT, L ME\*NASAS, MPBR  
ELT 005-07/13-20:59  
000001 000 CLASS  
000002 000 IN  
000003 000 DEF  
000004 000 END

MPBR  
CPMP, DTHCMP, ELAS, THCMP  
CPMP, DTHCMP, ELAS, THCMP

DHGG, P \*\*\*\*\* MPCU \*\*\*\*\*



\*\*\*\*\* MPCU \*\*\*\*\*

DELT,L ME\*NASAS,MPCU  
 ELT 005-07/13-20:59  
 000001 000 CLASS MPCU  
 000002 000 IN CPMP,OTHCMP,ELAS,THCMP  
 000003 000 DEF CPMP,OTHCMP,ELAS,THCMP  
 000004 000 END

QH06rP \*\*\*\*\* SCZ93 \*\*\*\*\*

\*\*\*\*\* SCZ93 \*\*\*\*\*

QELT,L ME\*NASAS,SCZ93  
ELT 005-07/13-20:59  
000001 000 CLASS SCZ93  
000002 000 IN AVGMT\*EMIT  
000003 000 DEF AVGMT\*EMIT  
000004 000 END

QHDG,P \*\*\*\*\* TBAL \*\*\*\*\*

\*\*\*\*\* TBAL \*\*\*\*\*

GELT,L ME\*NASAS,TBAL  
ELT 005-0713-20:59  
000001 000 CLASS TBAL  
000002 000 IN CPTB,DTHCTB,THCTB  
000003 000 DEF CPTB,DTHCTB,THCTB  
000004 000 END

SHDG,P \*\*\*\*\* TBBR \*\*\*\*\*

\*\*\*\*\* TBBR \*\*\*\*\*

QELT,L ME\*NASAS,TBBR  
ELT 005-07/13-20:59  
000001 000 CLASS TBBR  
000002 000 IN CPTB,DYHCTB,THCTB  
000003 000 DEF CPTB,DYHCTB,THCTB  
000004 000 END

QHDG,P \*\*\*\*\* TBCU \*\*\*\*\*

\*\*\*\*\* TBCU \*\*\*\*\*

GELT,L ME\*NASAS.TBCU  
ELT 005-07/13-20:59  
000001 000 CLASS TBCU  
000002 000 IN CPTB,DTHTCTB,THCTB  
000003 000 DEF CPTB,DTHTCTB,THCTB  
000004 000 END

9HDG\*P \*\*\*\*\* VER1 \*\*\*\*\*

\*\*\*\*\* VER1 \*\*\*\*\*

DELT,L ME\*NASAS,VER1  
ELT 005-07/13-20:59  
000001 000 IN  
000002 000 IN  
000003 000 END

CFSIL,FNAL,MPBR,TBAL,SCZ93  
MAIN

BHDG,N

BHDG,P \*\*\*\*\* CFFC43 \*\*\*\*\*

\*\*\*\*\* CFFC43 \*\*\*\*\*

MAP,RS ME\*NASAS.CFFC43  
 MAP 0023-07/13-20:53 -(0,)

- 1. NEW LIB ME\*NASAS.
- 2. CLASS CFFC43
- 3. IN BETA,CAPPA,CPF,HFL,RHOF,THCF,VISC
- 4. DEF BETA,CAPPA,CPF,HFL,RHOF,THCF,VISC
- 5. END

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED  
 \$(1) COMBINED ODD COUNTERS 000164 OCTAL 116 DECIMAL  
 \$(2) COMBINED EVEN COUNTERS 000105 OCTAL 69 DECIMAL  
 5 COMMONBLOCK B-BANK\$COMMON 000000 OCTAL 0 DECIMAL

COMBINED ELEMENTS				
BETA/CFFC43	1	000000	000016	0 000000 000007
CAPPA/CFFC43	1	000017	000030	2 BLANK\$COMMON
CPF/CFFC43	1	000031	000046	0 000010 000015
HFL/CFFC43	1	000047	000072	2 BLANK\$COMMON
RHOF/CFFC43	1	000073	000110	0 000016 000026
THCF/CFFC43	1	000111	000125	2 BLANK\$COMMON
VISC/CFFC43	1	000126	000163	0 000027 000042
				2 BLANK\$COMMON
				0 000043 000053
				2 BLANK\$COMMON
				0 000054 000063
				2 BLANK\$COMMON
				0 000064 000104
				2 BLANK\$COMMON

END OF COLLECTION - TIME 0.303 SECONDS

\*\*\*\*\* CFEC43 \*\*\*\*\*

QHDG:P \*\*\*\*\* CFFC75 \*\*\*\*\*

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\*\*\*\*\* CFFC75 \*\*\*\*\*

GMAP,RS ME\*NASAS.CFFC75  
 MAP 0023-07/13-20:59 -(0,)

- 1. NEW LIB ME\*NASAS.
- 2. CLASS CFFC75
- 3. IN BETA,CAPPA,CPF,HFL,RHOF,THCF,VISC
- 4. DEF BETA,CAPPA,CPF,HFL,RHOF,THCF,VISC
- 5. END

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED 116 DECIMAL  
 \$(1) COMBINED ODD COUNTERS 000164 OCTAL 69 DECIMAL  
 \$(2) COMBINED EVEN COUNTERS 000105 OCTAL 0 DECIMAL  
 3 COMMONBLOCK BLANK\$COMMON 000000 OCTAL

COMBINED ELEMENTS

BETA/CFFC75	1	000000	000016	0	000000	000007
CAPPA/CFFC75	1	000017	000030	2	BLANK\$COMMON	
CPF/CFFC75	1	000031	000046	2	000010	000015
HFL/CFFC75	1	000047	000072	2	BLANK\$COMMON	
RHOF/CFFC75	1	000073	000110	2	000016	000026
THCF/CFFC75	1	000111	000125	2	BLANK\$COMMON	
VISC/CFFC75	1	000126	000163	2	000027	000042
				2	BLANK\$COMMON	
				2	000043	000053
				2	BLANK\$COMMON	
				2	000054	000063
				2	BLANK\$COMMON	
				2	000064	000104
				2	BLANK\$COMMON	

END OF COLLECTION - TIME 0.203 SECONDS

\*\*\*\*\* CFFCT5 \*\*\*\*\*

CHDS.P \*\*\*\*\* CFHE \*\*\*\*\*

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\*\*\*\*\* CFHE \*\*\*\*\*

QMAP,RS ME\*NASAS5.CFHE  
 MAP 0023-07/13-20:59 -(0,)

- 1. NEW LIB ME\*NASAS5.
- 2. CLASS CFHE
- 3. IN BETA,CAPPA,CPF,HFL,PF,RHOF,THCF,VISC
- 4. DEF BETA,CAPPA,CPF,HFL,PF,RHOF,THCF,VISC
- 5. END

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED  
 \$(1) COMBINED ODD COUNTERS 000635 OCTAL 413 DECIMAL  
 \$(2) COMBINED EVEN COUNTERS 00267 OCTAL 183 DECIMAL  
 3 COMMONBLOCK BLANK\$COMMON 000000 OCTAL 0 DECIMAL

COMBINED ELEMENTS

BETA/CFHE	1	000000	000111	0	000000	000036
CAPPA/CFHE	1	000112	000200	2	BLANK\$COMMON	
CPF/CFHE	1	000201	000271	0	000037	000067
HFL/CFHE	1	000272	000415	2	BLANK\$COMMON	
PF/CFHE	1	000416	000467	0	000070	000120
RHOF/CFHE	1	000470	000566	2	BLANK\$COMMON	
THCF/CFHE	1	000567	000612	0	000121	000167
VISC/CFHE	1	000613	000634	2	BLANK\$COMMON	

END OF COLLECTION - TIME 0.315 SECONDS

\*\*\*\*\* CFHE \*\*\*\*\*

QHDGIP \*\*\*\*\* CFNAK \*\*\*\*\*

\*\*\*\*\* CFNAK \*\*\*\*\*

DMAP,RS ME\*NASAS.CFNAK  
 MAP 0023-07/13-20:59 -(0.)

- 1. NEW LIB ME\*NASAS.
- 2. CLASS CFNAK
- 3. IN BETA,CAPPA,CPF,HFL,PF,RHOF,THCF,VISC
- 4. DEF BETA,CAPPA,CPF,HFL,PF,RHOF,THCF,VISC
- 5. END

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED  
 \$(1) COMBINED ODD COUNTERS 000467 OCTAL  
 \$(2) COMBINED EVEN COUNTERS 000224 OCTAL  
 3 COMMONBLOCK BLANK\$COMMON 000000 OCTAL

311 DECIMAL  
 148 DECIMAL  
 0 DECIMAL

COMBINED ELEMENTS

BETA/CFNAK	1	000000	000016	0	000000	000007
				2	BLANK\$COMMON	
CAPPA/CFNAK	1	000017	000073	0	000010	000041
				2	BLANK\$COMMON	
CPF/CFNAK	1	000074	000177	0	000042	000073
				2	BLANK\$COMMON	
HFL/CFNAK	1	000200	000311	0	000074	000132
				2	BLANK\$COMMON	
PF/CFNAK	1	000312	000351	0	000133	000147
				2	BLANK\$COMMON	
RHOF/CFNAK	1	000352	000411	0	000150	000166
				2	BLANK\$COMMON	
THCF/CFNAK	1	000412	000437	0	000167	000204
				2	BLANK\$COMMON	
VISC/CFNAK	1	000440	000466	0	000205	000223
				2	BLANK\$COMMON	

END OF COLLECTION - TIME 0.273 SECONDS

\*\*\*\*\* CFNAK \*\*\*\*\*  
OHDS/P \*\*\*\*\* CFSIL \*\*\*\*\*

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\*\*\*\*\* CFSIL \*\*\*\*\*

MAP,RS ME\*NASAS.CFSIL  
 MAP 0023-07/13-20:59 -(0,)

1. NEW LIB ME\*NASAS.  
 2. CLASS CFSIL  
 3. IN BETA,CAPPA,CPF,HFL,PF,RHOF,THCF,VISC  
 4. DEF BETA,CAPPA,CPF,HFL,PF,RHOF,THCF,VISC  
 5. END

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED  
 \$(1) COMBINED ODD COUNTERS 001734 OCTAL  
 \$(2) COMBINED EVEN COUNTERS 000717 OCTAL  
 3 COMMONBLOCK BLANK\$COMMON 000000 OCTAL

988 DECIMAL  
 463 DECIMAL  
 0

COMBINED ELEMENTS

BETA/CFSIL	1	000000	000176	0	000000	000075
CAPPA/CFSIL	1	000177	000344	2	BLANK\$COMMON	000076
CPF/CFSIL	1	000345	000736	2	BLANK\$COMMON	000162
HFL/CFSIL	1	000737	001264	2	BLANK\$COMMON	000315
PF/CFSIL	1	001265	001446	2	BLANK\$COMMON	000447
RHOF/CFSIL	1	001447	001575	2	BLANK\$COMMON	000541
THCF/CFSIL	1	001576	001644	2	BLANK\$COMMON	000626
VISC/CFSIL	1	001645	001733	2	BLANK\$COMMON	000657
				2	BLANK\$COMMON	

END OF COLLECTION - TIME 0.469 SECONDS

\*\*\*\*\* CFSIL \*\*\*\*\*

QHDG,P \*\*\*\*\* FNAL \*\*\*\*\*



\*\*\*\*\* FNAL \*\*\*\*\*

QMAP,RS ME\*NASAS5.FNAL  
MAP 0023-07/13-20:59 -(0,)

- 1. NEW LIB ME\*NASAS5.
- 2. CLASS FNAL
- 3. IN CPFN,DTNCFN,THCFN
- 4. DEF CPFN,DTNCFN,THCFN
- 5. END

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED  
 \$ (1) COMBINED ODD COUNTERS 000202 OCTAL  
 \$ (2) COMBINED EVEN COUNTERS 000116 OCTAL  
 3 COMMONBLOCK BLANK\$COMMON 000000 .OCTAL

130 DECIMAL  
 78 DECIMAL  
 0 DECIMAL

COMBINED ELEMENTS

CPFN/FNAL	1	000000	000061	0	000000	000037
				2	BLANK\$COMMON	
DTHCFN/FNAL	1	000062	000121	0	000040	000056
				2	BLANK\$COMMON	
THCFN/FNAL	1	000122	000201	0	000057	000115
				2	BLANK\$COMMON	

END OF COLLECTION - TIME 0.171 SECONDS

ENDG,P \*\*\*\*\* FNBR \*\*\*\*\*

\*\*\*\*\* FNBR \*\*\*\*\*  
 QMAP,RS ME\*NASAS.FNBR  
 MAP 0023-07/13-20:59 -(0,)

1. NEW LIB ME\*NASAS.  
 2. CLASS FNBR  
 3. IN CPFN,DTHCFN,THCFN  
 4. DEF CPFN,DTHCFN,THCFN  
 5. END

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED  
 3(1) COMBINED ODD COUNTERS 000200 OCTAL 128 DECIMAL  
 3(2) COMBINED EVEN COUNTERS 000115 OCTAL 77 DECIMAL  
 3 COMMONBLOCK BLANKSCOMMON 000000 OCTAL 0 DECIMAL

COMBINED ELEMENTS

CPFN/FNBR	1	000000	000057	0	000000	000036
DTHCFN/FNBR	1	000060	000117	2	BLANKSCOMMON	
THCFN/FNBR	1	000120	000177	0	000037	000055
				2	BLANKSCOMMON	
				0	000056	000114
				2	BLANKSCOMMON	

END OF COLLECTION - TIME 0.169 SECONDS

SHDG,P \*\*\*\*\* FNCU \*\*\*\*\*

\*\*\*\*\* FNCU \*\*\*\*\*

QMAP,RS ME\*NASAS,FNCU  
MAP 0023-07/13-20:59 -(0,)

1. NEW LIB ME\*NASAS.  
 2. CLASS FNCU  
 3. IN CPFN,DTHCFN,THCFN  
 4. DEF CPFN,DTHCFN,THCFN  
 5. END

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED  
 \$(1) COMBINED ODD COUNTERS 000163 OCTAL 115 DECIMAL  
 \$(2) COMBINED EVEN COUNTERS 000102 OCTAL 66 DECIMAL  
 3 COMMONBLOCK BLANK\$COMMON 000000 OCTAL 0 DECIMAL

COMBINED ELEMENTS

CPFN/FNCU	1	000000	000050	0	000000	000030
DTHCFN/FNCU	1	000051	000105	0	BLANK\$COMMON	
THCFN/FNCU	1	000106	000162	0	000031	000045
				2	BLANK\$COMMON	
				0	000046	000101
				2	BLANK\$COMMON	

END OF COLLECTION - TIME 0.165 SECONDS

QHDG,P \*\*\*\*\* MPAL \*\*\*\*\*

\*\*\*\*\* MPAL \*\*\*\*\*

DMAP,RS ME\*NASAS,MPAL  
MAP 0023-07/13-20:59 -(0.)

- 1. NEW LIB ME\*NASAS.
- 2. CLASS MPAL
- 3. IN CPMP,DTHCMP,ELAS,THCMP
- 4. DEF CPMP,DTHCMP,ELAS,THCMP
- 5. END

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED

\$ (1)	COMBINED ODD COUNTERS	000263	OCTAL	179	DECIMAL	
\$ (2)	COMBINED EVEN COUNTERS	000156	OCTAL	110	DECIMAL	
3	COMMONBLOCK	BLANK\$COMMON	000000	OCTAL	0	DECIMAL

COMBINED ELEMENTS

CPMP/MPAL	1	000000	000061	0	000000	000037
				2	BLANK\$COMMON	
DTHCMP/MPAL	1	000062	000121	0	000040	000056
				2	BLANK\$COMMON	
ELAS/MPAL	1	000122	000202	0	000057	000116
				2	BLANK\$COMMON	
THCMP/MPAL	1	000203	000262	0	000117	000155
				2	BLANK\$COMMON	

END OF COLLECTION - TIME 0.206 SECONDS

DHDG,P \*\*\*\*\* MPBR \*\*\*\*\*

\*\*\*\*\* MPBR \*\*\*\*\*

QMAP,RS ME\*NASAS,MPBR  
 MAP 0023-07/13-20:59 -(0,)

1.NEW LIB VE\*NASAS.  
 2. CLASS MPBR  
 3. IN CPMP,DTHCMP,ELAS,THCMP  
 4. DEF CPMP,DTHCMP,ELAS,THCMP  
 5. END

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED  
 \$(1) COMBINED ODD COUNTERS 000261 OCTAL 177 DECIMAL  
 \$(2) COMBINED EVEN COUNTERS 000155 OCTAL 109 DECIMAL  
 3 COMMONBLOCK BLANK\$COMMON 000000 OCTAL 0 DECIMAL

COMBINED ELEMENTS

CPMP/MPBR	1	000000	000057	0	000000	000036
				2	BLANK\$COMMON	
DTHCMP/MPBR	1	000060	000117	0	000037	000055
				2	BLANK\$COMMON	
ELAS/MPBR	1	000120	000200	0	000056	000115
				2	BLANK\$COMMON	
THCMP/MPBR	1	000201	000260	0	000116	000154
				2	BLANK\$COMMON	

END OF COLLECTION - TIME 0.192 SECONDS

QHDG:P \*\*\*\*\* MPBU \*\*\*\*\*

QMAP,RS ME\*NASAS,MPCU  
 MAP 0023-07/13-20:59 -(0,)

- 1. NEW LIB ME\*NASAS.
- 2. CLASS MPCU
- 3. IN CPMP,DTHCMP,ELAS,THCMP
- 4. DEF CPMP,DTHCMP,ELAS,THCMP
- 5. END

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED  
 \$(1) COMBINED ODD COUNTERS 000244 OCTAL 164 DECIMAL  
 \$(2) COMBINED EVEN COUNTERS 000142 OCTAL 98 DECIMAL  
 3 COMMONBLOCK BLANK\$COMMON 000000 OCTAL 0 DECIMAL

COMBINED ELEMENTS				
CPMP/MPCU	1	000000	000050	0 000000 000030
DTHCMP/MPCU	1	000051	000105	2 BLANK\$COMMON
ELAS/MPCU	1	000106	000166	0 000031 000045
THCMP/MPCU	1	000167	000243	2 BLANK\$COMMON
				0 000046 000105
				2 BLANK\$COMMON
				0 000106 000141
				2 BLANK\$COMMON

END OF COLLECTION - TIME 0.187 SECONDS

QHDG,P \*\*\*\*\* SCZ93 \*\*\*\*\*

\*\*\*\*\* SCZ93 \*\*\*\*\*

QMAP,RS ME\*NASAS,SCZ93  
MAP 0023-07/13-20:59 -(0,)

1.	NE#	LIB	ME*NASAS.
2.	CLASS	SCZ93	
3.	IN	AVGEMT,EMIT	
4.	DEF	AVGEMT,EMIT	
5.	END		

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED		
\$ (1) COMBINED ODD COUNTERS	000156	OCTAL
\$ (2) COMBINED EVEN COUNTERS	000056	OCTAL
3 COMMONBLOCK AVGABS	000251	OCTAL
4 COMMONBLOCK BLANK\$COMMON	000000	OCTAL

COMBINED ELEMENTS

AVGEMT/SCZ93	1	000000	000120	0	000000	000037
EMIT/SCZ93	3	AVGABS		2	BLANK\$COMMON	
	1	000121	000155	0	000040	000055
				2	BLANK\$COMMON	

END OF COLLECTION - TIME 0.148 SECONDS

QHDG,P \*\*\*\*\* TBAL \*\*\*\*\*

\*\*\*\*\* TBAL \*\*\*\*\*

DMAPRS ME\*NASAS\*TBAL  
 MAP 0023-07/13-21:00 -(0,)

1. NEW LIB ME\*NASAS.  
 2. CLASS TBAL  
 3. IN CPTB,DTHCTB,THCTB  
 4. DEF CPTB,DTHCTB,THCTB  
 5. END

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED  
 \$(1) COMBINED ODD COUNTERS 000202 OCTAL 130 DECIMAL  
 \$(2) COMBINED EVEN COUNTERS 000116 OCTAL 78 DECIMAL  
 3 COMMONBLOCK BLANK\$COMMON 000000 OCTAL 0 DECIMAL

COMBINED ELEMENTS

CPTB/TBAL	1	000000	000061	0	000000	000037
DTHCTB/TBAL	1	000062	000121	0	BLANK\$COMMON	
THCTB/TBAL	1	000122	000201	0	000040	000056
				2	BLANK\$COMMON	
				0	000057	000115
				2	BLANK\$COMMON	

END OF COLLECTION - TIME 0.170 SECONDS

DM06:P \*\*\*\*\* TBRR \*\*\*\*\*



\*\*\*\*\* TBRR \*\*\*\*\*

QMAP,RS ME\*NASAS5,TBRR  
 MAP 0023-07/13-21:00 -(0,)

- 1. NEW LIB ME\*NASAS5.
- 2. CLASS TBRR
- 3. IN CPTB,DTHTCTB,THCTB
- 4. DEF CPTB,DTHTCTB,THCTB
- 5. END

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED  
 \$(1) COMBINED ODD COUNTERS 000200 OCTAL 128 DECIMAL  
 \$(2) COMBINED EVEN COUNTERS 000115 OCTAL 77 DECIMAL  
 3 COMMONBLOCK BLANK\$COMMON 000000 OCTAL 0 DECIMAL

COMBINED ELEMENTS

CPTB/TBRR	1	000000	000057	0	000000	000036
DTHTCTB/TBRR	1	000060	000117	0	2	BLANK\$COMMON
THCTB/TBRR	1	000120	000177	0	2	BLANK\$COMMON
				0	0	000037 000055
				0	0	BLANK\$COMMON
				0	2	000056 000114
				0	2	BLANK\$COMMON

END OF COLLECTION - TIME 0.167 SECONDS

ENDG,P \*\*\*\*\* TBCU \*\*\*\*\*

\*\*\*\*\* TBCU \*\*\*\*\*

MAP,RS ME\*NASAS.TBCU  
 MAP 0023-07/13-21:00 -(0,)

1. NEW LIB ME\*NASAS.  
 2. CLASS TBCU  
 3. IN CPTB,DTHTCTB,THCTB  
 4. DEF CPTB,DTHTCTB,THCTB  
 5. END

R-OPTION COLLECTION

LENGTHS OF LOCATION COUNTERS ASSIGNED  
 \$(1) COMBINED ODD COUNTERS 000163 OCTAL 115 DECIMAL  
 \$(2) COMBINED EVEN COUNTERS 000102 OCTAL 66 DECIMAL  
 3 COMMONBLOCK BLANK\$COMMON 000000 OCTAL 0 DECIMAL

COMBINED ELEMENTS

CPTB/TBCU	1	000000	000050	0	000000	000030
DTHTCTB/TBCU	1	000051	000105	2	BLANK\$COMMON	
THCTB/TBCU	1	000106	000162	2	BLANK\$COMMON	
				2	000046	000101
				2	BLANK\$COMMON	

END OF COLLECTION - TIME 0.165 SECONDS

SPREP ME\*NASAS.  
 FURPUR 023A-07/13-21:00

QHDG,P \*\*\*\*\* VER1 \*\*\*\*\*

MAP,S ME\*NASAS.VER1  
 MAP 0023-07/13-21:00 -(0,)

1. NEW LIB ME\*NASAS.  
 2. IN CFSIL\*FNAL,MPBR\*TBAL,SC293  
 3. IN MAIN  
 4. END

ADDRESS LIMITS 001000 052007 053000 116661  
 STARTING ADDRESS 046014  
 WORDS DECIMAL 21000 IBANK 18354 DBANK

	SEGMENT	YAIN	001000	052007	053000	116661
NFTVS\$/FOR	1	001000	001022			
NBF00\$/FOR	1	001023	001045	2	053000	055201
NRBLK\$/FOR	1	001046	001200	2	055202	055237
NBCCV\$/FOR	1	001201	001222			
NSWTC\$/FOR	1	001223	001257			
NBSL\$/FOR	1	001260	001312			
NUPJA\$/FOR	1	001313	001434			
NRLK\$/FOR	1	001435	001725	2	055240	055275
NFTCH\$/FOR	1	001726	002570	2	055276	055315
NINPT\$/FOR	1	002571	003104	2	055316	055321
NOTIN\$/FOR	1	003105	004073	2	055322	055346
NOUIT\$/FOR	1	004074	004752	2	055347	055423
NXVT\$/FOR	1	004753	005177	2	055424	055513
NIGR\$/FOR	1	005200	005352	2	055514	055620
NININ\$/FOR	1	005353	005563	2	055621	055632
NFCHK\$/FOR	1	005564	006445	2	055633	055771
NWFF\$/FOR	1	006446	006647	4	055772	056043
ERU\$/MISC	1			2	056044	056063
NFA\$/FOR	1	006650	006735	2	056064	056122
NEXP5\$/FOR	1	006736	007070	2	056123	056132
SINCO5\$/FOR	1	007071	007274	2	056133	056154
ATAN\$/FOR	1	007275	007534	2	056155	056206
NOSYM\$/FOR	1	007535	007614	2	056207	056213
NRWNS\$/FOR	1	007615	010011	2	056214	056225
NEXP6\$/FOR	1	010012	010510	2	056226	056277
NBRSP\$/FOR	1			2	056300	056326

\*\*\*\*\* VER1 \*\*\*\*\*

EXPS/FOR	1	010511	010600	2	056327	056347
ALOGS/FOR	1	010601	010720	2	056350	056410
NERRS/FOR	1	010721	011245	2	056411	056554
ASIVCOS\$/FOR	1	011246	011462	0	056555	056602
SORT\$/FOR	1	011463	011523	0	056603	056614
NLOUTS/FOR	1	011524	012606	2	056615	056652
NFOUTS/FOR	1	012607	013117	2	056653	056674
NFINPS/FOR	1	013120	013433	2	056675	056761
NIBUFS/FOR	1	013434	013476	2	056762	056762
NERS\$/FOR	1	013477	013560	2	056763	057116
NOBUFS/FOR	1	013561	013625	2	057117	057300
NLINPS/FOR	1	-013626	015341	0	057301	057331
TIRS/TECH	1	-015342	016026	0	057332	057611
NCLOSS\$/FOR	1	016027	016175	2	057612	057643
TTIP	1	016176	016255	0	057644	057744
EFFICY	1	016256	016436	0	BLANK\$COMMON	
DEFNT	1	016437	016516	0	057745	060075
EXITAV	1	016517	016626	2	BLANK\$COMMON	
	3	GRD		0	060076	060120
MTXINV	5	ABRSST		2	BLANK\$COMMON	
	1	016627	017034	0	060121	060150
TRMAYX	1	GRD		2	BLANK\$COMMON	
	1	017035	017331	0	060151	060207
	3	GRD		2	BLANK\$COMMON	
ABSORB	1	017332	020131	0	ABRSST	
	3	GRD		0	060261	060375
INTERP	5	AVGABS		2	BLANK\$COMMON	
	1	020132	020357	4	ABRSST	
NUS	1	020360	021221	0	060376	060460
REFF	1	021222	021610	2	BLANK\$COMMON	
CPAIR	1	021611	021720	2	060461	060574
TNH	1	021721	022004	0	BLANK\$COMMON	
ENTAIR	1	022005	022231	2	BLANK\$COMMON	
AT40S	1	022232	022512	0	060575	060757
ALTVEL	1	022513	022721	2	BLANK\$COMMON	
RAXF	1	022722	023761	0	060760	061014
TRNSPT	1	023762	024152	2	BLANK\$COMMON	
DRVLCN (COMMON BLOCK)				0	061015	061044
DRVLCM (COMMON BLOCK)				2	BLANK\$COMMON	
CNTLN	1	024153	024263	0	061045	061135
	3	DRVLCM		2	BLANK\$COMMON	
				0	061136	061346
				2	BLANK\$COMMON	
				0	061347	061364
				2	BLANK\$COMMON	
				0	061365	061450
				2	BLANK\$COMMON	
				0	061451	061510
				2	BLANK\$COMMON	
				0	061511	061511
				2	061512	061535
				0	061536	061560
				2	BLANK\$COMMON	
				4	DRVLCN	

DERIVL	1	024264	024542	0	061561	061633
FLCMP	3	024543	025050	2	BLANK\$COMMON	
FMINV	1	025051	025254	0	061634	063113
PDERIV	1	ADBH		2	BLANK\$COMMON	
TIPS	3	025255	025642	0	063114	063133
	3	ADBH		2	BLANK\$COMMON	
	3	ADBH		0	063134	063243
NUSA	1	025643	025745	4	BLANK\$COMMON	
	1			4	GIN	
SHADE	1	025746	026051	0	063244	063267
	1			2	BLANK\$COMMON	
FINT	1	026052	026162	0	063270	063312
	1			2	BLANK\$COMMON	
YINT	1	026163	026477	0	063313	063341
	1			2	BLANK\$COMMON	
D2DX2	1	026500	026662	0	063342	063477
	1			2	BLANK\$COMMON	
DDX	1	026663	027035	0	063500	063545
	1			2	BLANK\$COMMON	
GRAD	1	027036	030050	0	063546	063613
	3	GRD		2	BLANK\$COMMON	
	5	ABRST		0	063614	064712
CONVEC	7	AVGABS		2	BLANK\$COMMON	
	1	030051	030631	0	064713	065056
	3	VELALT		2	BLANK\$COMMON	
	5	SRTCNV		4	CNV	
	7	SSF		2	GRD	
POLY	1	030632	030675	6	SSF	
	1			0	065057	065073
RKS	1	030676	031735	0	BLANK\$COMMON	
SHAPEF	1	031736	032772	2	065074	065157
	3	GRD		0	BLANK\$COMMON	
DEFINT	1	032773	033217	2	BLANK\$COMMON	
	1			0	065160	065331
FLCMP (COMMON BLOCK)	1	033220	033464	2	BLANK\$COMMON	
FLSTRT	3	DVCMFL		0	065400	065423
	3			2	065424	065656
TK	1	033465	033731	4	BLANK\$COMMON	
	1			4	FLCMP	
GINCID	1	033732	034760	0	065657	065740
	3	GIN		2	BLANK\$COMMON	
	3			0	065741	066b25
ADIABH	1	034761	035502	2	BLANK\$COMMON	
	3	ADBH		4	TC	
TCALC	1	035503	036354	0	066626	067276
	3	ADBH		2	BLANK\$COMMON	
	5	GIN		0	067277	070166
SRTCNV (COMMON BLOCK)	1	036355	040434	2	BLANK\$COMMON	
CNTLM	3	SRTCNV		4	TC	
	5	GIN		0	070167	070171
	7	FLDINL		2	070172	073614
				4	BLANK\$COMMON	
				6	GRD	
					GEOM	

\*\*\*\*\* VER1 \*\*\*\*\*

DERIVM	1	04035 043014	8	DVCMFL
	0	DVM	0	073615 074675
	2	FLDINL	2	BLANK\$COMMON
	4		4	GIN
	6		6	DVCMFL
AVGABS (COMMON BLOCK)				074676 075146
DVCMFL (COMMON BLOCK)				075147 075150
FLDINL (COMMON BLOCK)				075151 075627
GEOM (COMMON BLOCK)				075630 075647
SSF (COMMON BLOCK)				075650 075657
ABSRST (COMMON BLOCK)				075660 076460
TC (COMMON BLOCK)				076461 076522
GIN (COMMON BLOCK)				076523 100332
GRD (COMMON BLOCK)				100333 104253
CNV (COMMON BLOCK)				104254 104203
VELALT (COMMON BLOCK)				104264 105415
DVA (COMMON BLOCK)				105416 105417
ADBH (COMMON BLOCK)				105420 105771
BLANK\$COMMON (COMMON BLOCK)				105772 107106
CFSIL	1	043015 044750	2	107107 110025
	3	BLANK\$COMMON	2	110026 110143
FNAL	1	044751 045152	2	
	3	BLANK\$COMMON	2	110144 110320
MPBR	1	045153 045433	2	
	3	BLANK\$COMMON	2	110321 110436
TBAL	1	045434 045635	2	
	3	BLANK\$COMMON	2	110437 110514
SCZ93	1	045636 046013	4	BLANK\$COMMON
	3	AVGABS	0	110515 116661
MAIN	1	046014 052007	2	BLANK\$COMMON
	3	ADBH	4	DVM
	5	VELALT	6	CNV
	7	GRD	8	GIN
	9	TC	10	ABSRST
	11	SSF	12	GEOM
	13	FLDINL	14	DVCMFL

SYS\$RLIB\$. LEVEL 63  
END OF COLLECTION - TIME 4.525 SECONDS

QHD6.P \*\*\*\*\* TABLE OF CONTENTS \*\*\*\*\*L.1

QPRINT ME\*NASAS,  
FURPUR 023A-07/13-21:00

ME\*NASAS ELEMENT TABLE

D	NAME	VERSION	TYPE	DATE	TIME	SEQ #	SIZE-PR	TEXT (CYCLE WORD)	PSRMODE	LOCATION	
	ABSORB		FOR SYMB	09 MAY 72	16:50:13	1	5	0	1	1792	
	ADIABH		FOR SYMB	02 MAY 72	15:31:15	2	24	5	0	1813	
	ALTVEL		FOR SYMB	09 MAY 72	16:50:24	3	7	5	0	1837	
	ATMOS		FOR SYMB	09 MAY 72	16:50:26	4	24	5	0	1844	
	AVGEMT		FOR SYMB	09 MAY 72	15:48:01	5	8	5	0	1868	
	BETA	SCZ93	FOR SYMB	09 MAY 72	16:43:19	6	3	5	0	1876	
	BETA	CFFC*3	FOR SYMB	09 MAY 72	16:44:12	7	3	5	0	1879	
	BETA	CFFC75	FOR SYMB	09 MAY 72	16:44:12	8	6	5	0	1882	
	BETA	CFHE	FOR SYMB	09 MAY 72	16:40:41	9	3	5	0	1888	
	BETA	CFNAK	FOR SYMB	09 MAY 72	16:42:30	10	8	5	0	1891	
	CAPPA	CFSIL	FOR SYMB	30 JUN 72	15:44:49	11	3	5	0	1899	
	CAPPA	CFFC*3	FOR SYMB	09 MAY 72	16:43:25	12	3	5	0	1902	
	CAPPA	CFFC75	FOR SYMB	09 MAY 72	16:44:34	13	5	5	0	1905	
	CAPPA	CFHE	FOR SYMB	09 MAY 72	16:40:48	14	5	5	0	1910	
	CAPPA	CFNAK	FOR SYMB	09 MAY 72	16:42:37	15	5	5	0	1915	
	CAPPA	CFSIL	FOR SYMB	30 JUN 72	15:44:52	16	1	5	0	1922	
	CFFC*3	MAP SYMB	MAP SYMB	10 MAY 72	08:56:52	17	1	5	0	1923	
	CFFC75	MAP SYMB	MAP SYMB	10 MAY 72	08:57:18	18	1	5	0	1924	
	CFHE	MAP SYMB	MAP SYMB	10 MAY 72	08:58:15	19	1	5	0	1925	
	CFNAK	MAP SYMB	MAP SYMB	10 MAY 72	08:58:40	20	1	5	0	1926	
	CFSIL	MAP SYMB	MAP SYMB	10 MAY 72	08:59:02	21	77	5	0	1927	
	CNTLM	FOR SYMB	FOR SYMB	12 MAY 72	09:06:54	22	12	5	0	2004	
	CNTLN	FOR SYMB	FOR SYMB	09 MAY 72	16:49:43	23	28	5	0	2016	
	CONVEC	FOR SYMB	FOR SYMB	09 MAY 72	16:51:04	24	6	5	0	2044	
	CPAIR	FOR SYMB	FOR SYMB	09 MAY 72	16:50:38	25	3	5	0	2050	
	CPF	CFFC*3	FOR SYMB	09 MAY 72	16:43:31	26	3	5	0	2053	
	CPF	CFFC75	FOR SYMB	09 MAY 72	16:44:44	27	5	5	0	2056	
	CPF	CFHE	FOR SYMB	09 MAY 72	16:40:55	28	5	5	0	2061	
	CPF	CFNAK	FOR SYMB	09 MAY 72	16:42:43	29	13	5	0	2066	
	CPF	CFSIL	FOR SYMB	30 JUN 72	15:44:54	30	5	5	0	2079	
	CPFN	FNAL	FOR SYMB	09 MAY 72	16:47:50	31	5	5	0	2084	
	CPFN	FNBR	FOR SYMB	09 MAY 72	16:48:28	32	4	4	5	0	2089
	CPFN	FNCU	FOR SYMB	09 MAY 72	16:48:13	33	5	5	0	2093	
	CPMP	MPAL	FOR SYMB	09 MAY 72	16:48:38	34	5	5	0	2098	
	CPMP	MPBR	FOR SYMB	09 MAY 72	16:49:21	35	5	5	0	2103	
	CPMP	WPCU	FOR SYMB	09 MAY 72	16:48:52	36	4	4	5	0	2107
	CPTB	TBAL	FOR SYMB	09 MAY 72	16:46:59	37	5	5	0	2112	
	CPTB	TBBR	FOR SYMB	09 MAY 72	16:47:32	38	4	4	5	0	2117
	CPTB	TBCU	FOR SYMB	09 MAY 72	16:47:18	39	24	4	5	0	2121
	DATA		ELT SYMB	09 MAY 72	14:02:46	40	8	5	0	2145	
	DDX		FOR SYMB	09 MAY 72	16:51:51	41	6	5	0	2153	
	DEFINT		FOR SYMB	09 MAY 72	16:52:40	42	3	5	0	2159	
	DEFIT		FOR SYMB	09 MAY 72	16:52:04	43	18	5	0	2162	
	DERIVL		FOR SYMB	11 MAY 72	15:27:24	44	61	5	0	2180	
	DERIVM		FOR SYMB	09 MAY 72	16:52:10	45	4	5	0	2241	
	DTHCFN	FNAL	FOR SYMB	09 MAY 72	16:47:46	46	4	4	5	0	2245
	DTHCFN	FNBR	FOR SYMB	09 MAY 72	16:48:23	47	4	4	5	0	2249
	DTHCFN	WPCU	FOR SYMB	09 MAY 72	16:48:10	48	4	4	5	0	2253
	DTHCMP	MPAL	FOR SYMB	09 MAY 72	16:48:35	49	4	4	5	0	2257
	DTHCMP	MPBR	FOR SYMB	09 MAY 72	16:49:13	50	4	4	5	0	2261
	DTHCMP	WPCU	FOR SYMB	09 MAY 72	16:48:49	51	4	4	5	0	2265
	DTHCTB	TBAL	FOR SYMB	09 MAY 72	16:46:53						

\*\*\*\*\* TABLE OF CONTENTS \*\*\*\*\*

DTHTB	TBTR	FOR SYMB	09 MAY 72	16:47:28	52	4	5	0	1	2269
DTHCTB	TBCU	FOR SYMB	09 MAY 72	16:47:15	53	4	5	0	1	2273
D2DX2		FOR SYMB	09 MAY 72	16:51:54	54	9	5	0	1	2277
EFFICY		FOR SYMB	23 MAR 72	16:32:42	55	8	5	0	1	2286
ELAS	MPAL	FOR SYMB	09 MAY 72	16:48:42	56	5	5	0	1	2294
ELAS	MPBR	FOR SYMB	09 MAY 72	16:49:27	57	5	5	0	1	2299
ELAS	MPCU	FOR SYMB	09 MAY 72	16:48:54	58	5	5	0	1	2304
EMIT	SCZ93	FOR SYMB	09 MAY 72	15:47:54	59	4	5	0	1	2309
ENTAIR		FOR SYMB	09 MAY 72	16:50:46	60	9	5	0	1	2313
EXITAV		FOR SYMB	09 MAY 72	16:50:10	61	7	5	0	1	2322
FINT		FOR SYMB	09 MAY 72	16:52:43	62	4	5	0	1	2329
FLSTRY		FOR SYMB	11 MAY 72	15:48:02	63	27	5	0	1	2333
FMINV		FOR SYMB	23 MAR 72	16:34:18	64	15	5	0	1	2360
FNAL		MAP SYMB	10 MAY 72	08:59:59	65	1	5	0	1	2375
FNBR		MAP SYMB	10 MAY 72	09:00:26	66	1	5	0	1	2376
FNCU		MAP SYMB	10 MAY 72	09:00:51	67	1	5	0	1	2377
HFL	CFFC43	FOR SYMB	09 MAY 72	16:40:25	68	3	5	0	1	2378
HFL	CFFC75	FOR SYMB	09 MAY 72	16:45:21	69	3	5	0	1	2381
HFL	CFHE	FOR SYMB	09 MAY 72	16:40:58	70	6	5	0	1	2384
HFL	CFNAK	FOR SYMB	09 MAY 72	16:42:49	71	5	5	0	1	2390
HFL	CFSIL	FOR SYMB	30 JUN 72	15:44:56	72	12	5	0	1	2395
INTERP		FOR SYMB	09 MAY 72	16:52:45	73	7	5	0	1	2407
MAIN		FOR SYMB	09 MAY 72	15:58:27	74	197	5	0	1	2414
MPAL		MAP SYMB	10 MAY 72	09:02:03	75	1	5	0	1	2611
MPBR		MAP SYMB	10 MAY 72	09:02:27	76	1	5	0	1	2612
MPCU		MAP SYMB	10 MAY 72	09:03:11	77	1	5	0	1	2613
MIXINV		FOR SYMB	09 MAY 72	16:51:59	78	7	5	0	1	2614
NUS		FOR SYMB	09 MAY 72	16:52:39	79	20	5	0	1	2621
NUSA		FOR SYMB	20 APR 72	13:23:20	80	3	5	0	1	2641
PUERIV		FOR SYMB	02 MAY 72	15:35:16	81	7	5	0	1	2644
PF	CFHE	FOR SYMB	09 MAY 72	16:41:10	82	4	5	0	1	2651
PF	CFNAK	FOR SYMB	09 MAY 72	16:42:58	83	3	5	0	1	2655
PF	CFSIL	FOR SYMB	30 JUN 72	15:45:03	84	8	5	0	1	2658
POLY		FOR SYMB	09 MAY 72	16:51:39	85	3	5	0	1	2666
QINCID		FOR SYMB	09 MAY 72	16:26:12	86	25	5	0	1	2669
GRAD		FOR SYMB	09 MAY 72	16:50:18	87	33	5	0	1	2694
REFP		FOR SYMB	09 MAY 72	16:50:58	88	21	5	0	1	2727
RHOF		FOR SYMB	09 MAY 72	16:43:09	89	3	5	0	1	2748
RHOF	CFFC43	FOR SYMB	09 MAY 72	16:44:04	90	3	5	0	1	2751
RHOF	CFFC75	FOR SYMB	09 MAY 72	16:40:33	91	5	5	0	1	2754
RHOF	CFHE	FOR SYMB	09 MAY 72	16:42:23	92	4	5	0	1	2759
RHOF	CFNAK	FOR SYMB	09 MAY 72	16:42:47	93	7	5	0	1	2763
RHOF	CFSIL	FOR SYMB	30 JUN 72	15:44:47	94	73	5	0	1	2770
RKSF		FOR SYMB	09 MAY 72	16:51:28	95	73	5	0	1	2770
SCZ93		FOR SYMB	09 MAY 72	16:51:18	96	1	5	0	1	2843
SHADE		MAP SYMB	10 MAY 72	09:06:45	97	1	5	0	1	2916
SHAPEF		FOR SYMB	20 APR 72	14:28:06	98	3	5	0	1	2917
TBAL		FOR SYMB	09 MAY 72	16:49:51	99	24	5	0	1	2920
TBBR		MAP SYMB	10 MAY 72	09:41:22	100	1	5	0	1	2944
TBCU		MAP SYMB	10 MAY 72	09:41:33	101	1	5	0	1	2945
TCALC		MAP SYMB	10 MAY 72	09:05:22	102	20	5	0	1	2946
THCF	CFFC43	FOR SYMB	02 MAY 72	15:36:13	103	23	5	0	1	2947
THCF	CFFC75	FOR SYMB	09 MAY 72	16:43:49	104	3	5	0	1	2967
THCF	CFHE	FOR SYMB	09 MAY 72	16:46:38	105	4	5	0	1	2970
THCF	CFNAK	FOR SYMB	09 MAY 72	16:41:06	106	4	5	0	1	2973
THCF	CFNAK	FOR SYMB	09 MAY 72	16:42:55	107	4	5	0	1	2977
THCF	CFSIL	FOR SYMB	30 JUN 72	15:45:00	108	5	5	0	1	2981
THCFN	FNAL	FOR SYMB	09 MAY 72	16:47:40	108	5	5	0	1	2986



THCFN	FNBR	FOR SYMB	09 MAY 72	16:48:17	109	2	5	5	0	1	2991
THCFN	FNBR	FOR SYMB	09 MAY 72	16:48:17	109	2	5	5	0	1	2991
THCFN	FNBR	FOR SYMB	09 MAY 72	16:47:57	110	2	5	5	0	1	2996
THCMP	MPAL	FOR SYMB	09 MAY 72	16:48:33	111	5	5	5	0	1	3001
THCMP	MPBR	FOR SYMB	09 MAY 72	16:48:57	112	5	5	5	0	1	3006
THCMP	MPCU	FOR SYMB	09 MAY 72	16:48:44	113	5	5	5	0	1	3011
THCTB	TBAL	FOR SYMB	09 MAY 72	16:46:49	114	5	5	5	0	1	3016
THCTB	TBBR	FOR SYMB	09 MAY 72	16:47:24	115	5	5	5	0	1	3021
THCTB	TBCU	FOR SYMB	09 MAY 72	16:47:04	116	5	5	5	0	1	3026
TK		FOR SYMB	09 MAY 72	16:51:09	117	5	5	5	0	1	3031
TN		FOR SYMB	09 MAY 72	16:50:50	118	4	5	5	0	1	3038
TRMATX		FOR SYMB	09 MAY 72	16:50:04	119	11	5	5	0	1	3042
TRNSPT		FOR SYMB	09 MAY 72	16:52:30	120	11	5	5	0	1	3053
TTIP		FOR SYMB	23 MAR 72	16:33:46	121	5	5	5	0	1	3064
TTIPS		FOR SYMB	02 MAY 72	15:37:06	122	15	5	5	0	1	3069
VERI		MAP SYMB	10 MAY 72	16:52:30	123	1	5	5	0	1	3084
VISC	CFFC43	FOR SYMB	09 MAY 72	16:43:41	124	4	5	5	0	1	3085
VISC	CFFC75	FOR SYMB	09 MAY 72	16:45:47	125	4	5	5	0	1	3089
VISC	CFHE	FOR SYMB	09 MAY 72	16:41:03	126	3	5	5	0	1	3093
VISC	CFNAK	FOR SYMB	09 MAY 72	16:42:52	127	4	5	5	0	1	3096
VISC	CFNSIL	FOR SYMB	30 JUN 72	15:44:58	128	5	5	5	0	1	3100
YINT		FOR SYMB	09 MAY 72	16:28:50	129	10	5	5	0	1	3105
ABSORB		RELOCATABLE	13 JUL 72	20:52:57	130	22	22	22	0	1	3115
ADIABH		RELOCATABLE	13 JUL 72	20:53:00	131	27	27	27	0	1	3139
ALIVEL		RELOCATABLE	13 JUL 72	20:53:02	132	7	7	7	0	1	3168
ATMOS		RELOCATABLE	13 JUL 72	20:53:04	133	1	1	1	0	1	3176
AVGEMT	SC297	RELOCATABLE	13 JUL 72	20:53:06	134	1	6	6	0	1	3193
BETA	CFFC43	RELOCATABLE	13 JUL 72	20:53:09	135	1	2	2	0	1	3200
BETA	CFFC75	RELOCATABLE	13 JUL 72	20:53:10	136	1	2	2	0	1	3203
BETA	CFHE	RELOCATABLE	13 JUL 72	20:53:12	137	1	5	5	0	1	3206
BETA	CFNAK	RELOCATABLE	13 JUL 72	20:53:14	138	1	2	2	0	1	3212
BETA	CFNSIL	RELOCATABLE	13 JUL 72	20:53:16	139	1	10	10	0	1	3215
CAPPA	CFFC43	RELOCATABLE	13 JUL 72	20:53:18	140	1	1	1	0	1	3226
CAPPA	CFFC75	RELOCATABLE	13 JUL 72	20:53:21	141	1	1	1	0	1	3228
CAPPA	CFHE	RELOCATABLE	13 JUL 72	20:53:24	142	1	4	4	0	1	3230
CAPPA	CFNAK	RELOCATABLE	13 JUL 72	20:53:28	143	1	4	4	0	1	3235
CAPPA	CFNSIL	RELOCATABLE	13 JUL 72	20:54:16	144	1	8	8	0	1	3240
CNTLM		RELOCATABLE	13 JUL 72	20:54:25	145	3	78	78	0	1	3249
CNTLN		RELOCATABLE	13 JUL 72	20:54:31	146	2	5	5	0	1	3330
CONVEC		RELOCATABLE	13 JUL 72	20:54:37	147	3	22	22	0	1	3337
CPAIR		RELOCATABLE	13 JUL 72	20:54:40	148	1	6	6	0	1	3362
CPF	CFFC43	RELOCATABLE	13 JUL 72	20:54:42	149	1	2	2	0	1	3369
CPF	CFFC75	RELOCATABLE	13 JUL 72	20:54:45	150	1	2	2	0	1	3372
CPF	CFHE	RELOCATABLE	13 JUL 72	20:54:46	151	1	5	5	0	1	3375
CPF	CFNAK	RELOCATABLE	13 JUL 72	20:54:49	152	1	5	5	0	1	3381
CPF	CFNSIL	RELOCATABLE	13 JUL 72	20:54:53	153	1	17	17	0	1	3387
CPFN	FNAL	RELOCATABLE	13 JUL 72	20:54:56	154	1	5	5	0	1	3405
CPFN	FNBR	RELOCATABLE	13 JUL 72	20:54:58	155	1	5	5	0	1	3411
CPFN	FNCU	RELOCATABLE	13 JUL 72	20:55:00	156	1	4	4	0	1	3417
CPMP	MPAL	RELOCATABLE	13 JUL 72	20:55:02	157	1	5	5	0	1	3422
CPMP	MPBR	RELOCATABLE	13 JUL 72	20:55:07	158	1	5	5	0	1	3428
CPMP	MPCU	RELOCATABLE	13 JUL 72	20:55:10	159	1	5	5	0	1	3434
CPTB	TBAL	RELOCATABLE	13 JUL 72	20:55:05	160	1	5	5	0	1	3439
CPTB	TBBR	RELOCATABLE	13 JUL 72	20:55:12	161	1	5	5	0	1	3445
CPTB	TBCU	RELOCATABLE	13 JUL 72	20:55:14	162	1	4	4	0	1	3451
DDX		RELOCATABLE	13 JUL 72	20:55:16	163	1	7	7	0	1	3456
DEFINT		RELOCATABLE	13 JUL 72	20:55:21	164	1	9	9	0	1	3464
DEFNT		RELOCATABLE	13 JUL 72	20:55:24	165	1	3	3	0	1	3474





