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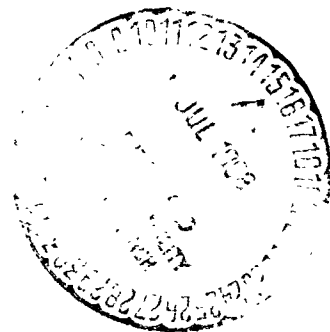
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VOYAGER MARS PLANETARY QUARANTINE -
BASIC MATH MODEL REPORT

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4800 OAK GROVE DRIVE
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TABLE OF CONTENTS

<u>Section</u>		<u>Page</u>
	INTRODUCTION	v
1	APPROACH TO MATH MODEL	1-1
2	MECHANIZATION	2-1
3	ENTRY	3-1
4	SENSITIVITY	4-1

INTRODUCTION

The prime goal of the Planetary Quarantine Subtask under the Voyager Phase 1A Task C Study is to show the effect of the Planetary Quarantine requirements on the Voyager mission and its elements.

Figure I is a simplified work flow diagram showing the major Planetary Quarantine subtasks and their interrelationships. Activities performed on this contract are being documented in bimonthly progress reports and a separate series of technical reports and memos. The present report presents the activities to date under the Basic Math Model Development Subtask. Section 1 discusses the basic questions involved and the selected approach to be used. Section 2 describes the mechanization of the basic concepts. Two previous reports, VOY-C2-TR7 and VOY-C2-TR4, presented work done in the basic parametric analysis subtasks for Orbit Mechanics and Entry Analysis. Section 3 of the present report presents the work performed in taking the basic entry results and converting them to a form suitable for incorporation into the math model computer programs. Section 4 of this report gives an example of how the math model is used in performing sensitivity studies, the next major subtask in the Planetary Quarantine Study. Appendix A gives a complete listing of the programs developed for this task.

Table I lists the reports issued to date on the Planetary Quarantine Task.

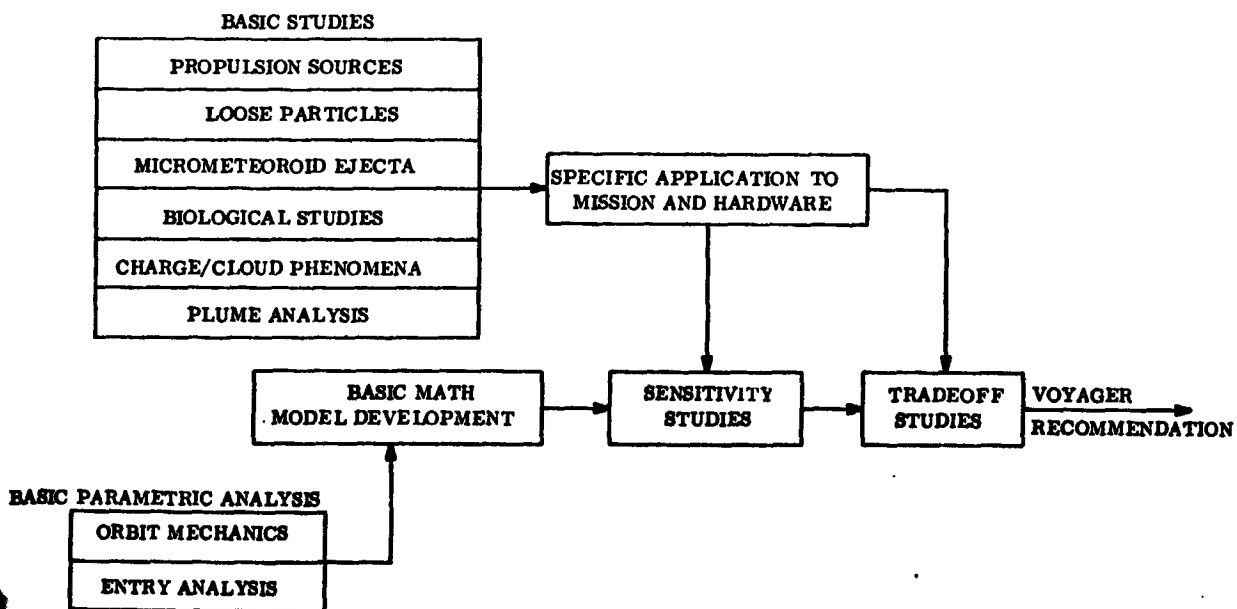


Figure I. Planetary Quarantine Task, Simplified Work Flow Diagram

Table I. Report Index for Voyager Planetary Quarantine Task C

VOY C2	Title	Author	Date
TR 1	On the Distribution of Density at Orbital Altitudes in the Martian Atmosphere.	Vachon	15 August 1966
TR 2	Prelim. Assessment of the Micrometeoroid Phenomena.	Good	July 1966
TR 3	Influence of Space Vehicle Charge and Plasma Field on the Quarantine of the Planet Mars.	McKee	1 September 1966
TR 4	Voyager Mars Planetary Quarantine Particle Burnup Study.	Parker Beeger Burrows	15 September 1966
TR 5	An Investigation into the Feasibility of Conducting an Experiment To Determine the Effects of Rocket Combustion on the Viability of Microorganisms.	Oberta	23 September 1966
TR 6	An Approximate Plume Analysis for the Voyager, Task C, Planetary Quarantine Study.	Hamel	30 September 1966
Tr 7	Voyager Mars Quarantine-Ejected Particle Trajectory Study.	Korenstein	30 November 1966
TM-1	Preliminary Combinatorial Probability Model for the Voyager Quarantine Problem (Phases 1, 2, 3).	T. Green	October 1966
TM-2	Voyager Mars P. Q. Thermal Kill of Bacteria during Mars Entry.	M. Martin	October 1966
TM-3	Loose Particle Investigation - An Evaluation of the Problem.	R. Waite	October 1966
TM-4	Micrometeoroid Effects - Analytical Studies Status Report - September 30, 1966.	R. Good	October 1966
TM-5	Voyager P. Q. Literature Search: Cold Gas Systems.	J. Mason	October 1966
TM-6	Preliminary Bio Burden Cata - Voyager P. Q.	M. Koesterer	October 1966
TM-7	Combustion Lethality Experiment - Status Report.	Oberta	November 1966
TM-8	Radiation Effects on the Viability of Microorganisms.	Peterson Koesterer	December 1966
TM-9	CLE-integ. Dev. Test Plan	Oberta Gillis Landry	December 1966
TM-10	Micrometeoroid Studies - Status Report, 12/66	Good Behringer Nayer	December 1966
TM-11	Micrometeoroid Simulation Experimental Studies - Status Report, Jan. 1967.	Koesterer Behringer Semon Nayer	January 1967
TM-12	Cold Gas ACS Experimental Program - Status Report January 1967.	Mason	January 1967
TM-13	Loose Particle Investigation - Status Report, January 1967.	Jones Rosta Nayer	January 1967

SECTION 1
APPROACH TO MATH MODEL
by
G. E. Ingram

TABLE OF CONTENTS

Section	Page
1	1-1
1.1 Rationale	1-1
1.2 Computational Problems	1-6
1.3 Implementation	1-10
1.4 Computer Program Development	1-17

LIST OF ILLUSTRATIONS

Figure	Page
1-1	1-2
1-2	1-3
1-3	1-4
1-4	1-5
1-5	1-6
1-6	1-7
1-7	1-8
1-8	1-9
1-9	1-11
1-10	1-12
1-11	1-14
1-12	1-14
1-13	1-15

SECTION 1
APPROACH TO MATH MODEL

1.1 RATIONALE

1.1.1 STATEMENT OF PROBLEM

The problem of determining the probability of contaminating Mars before time T, as a result of a Voyager mission, is essentially one of identifying all possible contamination sources associated with the "Voyager hardware" (launch vehicle, spacecraft, lander, etc.) and describing the various mechanisms that will cause viable organisms to reach the surface of Mars. The events of interest, therefore, can be described generically as follows---

One or more viable organisms launched from earth on Voyager hardware are placed on an impact trajectory to the surface of Mars using some mechanism and survive all potential kill mechanisms (e. g. , U. V. , atmospheric entry heating, etc.) and arrive, survive and spread on the surface of Mars before time T. The probability of the occurrence of all such events, then, is the probability of contaminating Mars before time T as a result of the missions.

Figure 1-1 illustrates the approach being used in matrix form. The rows of the matrix enumerate all possible sources of contamination while the columns of the matrix are descriptive of how particles find their way to the surface of Mars. For purposes of illustration, only four sources of contamination are listed.

ROUTE TO MARS
→

SOURCE OF CONTAMINATION	1	2	3	4	5	6	7	8	9	10	11
	INITIAL LOADING - V. C.	SURVIVE DURING TRIP	EJECTION PROCESS	TRANSPORT PROCESS	SURVIVE DIE-OFF	SURVIVE VACUUM	SURVIVE U. V.	SURVIVE OTHER SOLAR RADIATION	SURVIVE ENTRY HEATING	SURVIVE MARS ENVIRONMENT	NUMBER V. O.'S - MARS SURFACE PRIOR TO TIME T
ATTITUDE CONTROL GAS SYSTEM											
ORBIT INSERTION ENGINE											
LOOSE PARTICLES											
MICRO-METEOROID EJECTA											

Figure 1-1. Math Model Format

To determine how viable organisms might find their way to the surface of Mars, one must first consider the initial loading on the vehicle, second the transport process that the particles undergo in arriving at the surface of Mars, and third the potential kill mechanisms that the viable organisms are subjected to enroute to Mars. Column 1 calls for input data on the initial loading of viable organisms on the vehicle. Column 2 calls for data describing the probability that the viable organisms initially on the vehicle will survive during the trip before the time of ejection. Data describing the manner in which the particles are ejected from the vehicle is called for in Column 3. Column 4 describes the process by which the ejected particles find their way to Mars. Columns 5 through 10 call for data on the probability that the organisms will be killed enroute to Mars.

Finally after performing the operations indicated by Columns 1 through 10 on all of the sources of contamination indicated by the rows, the number of viable organisms that reach the surface of Mars and survive before time T is given in Column 11. Column 11 then is totalled for all possible sources of contamination giving the total number of viable organisms that reach the surface of Mars and survive before time T.

1.1.2 INPUT

Figure 1-2 illustrates the flow of information by considering some important elements of the problem, ---the initial loading of viable organisms, the transport process, the Mars atmosphere entry heating kill mechanism, and finally the number of viable organisms arriving at the surface of Mars and surviving before time T. Consider first the initial loading of viable organisms on the vehicle. In what terms might the initial loading be given? It might be stated as an average number of the several measurements of the loading. A more conservative approach would be to state it as some upper or maximum value of several measurements of the loading. A realistic description would be to describe this number as several ranges of values, each with an associated probability. Figure 1-3 graphically shows this type description. Note from the figure that the initial loading could be small in number, that is near 0, or it could be very large, as high as say one million. The probability however, that the number is very small, or very large, is a small probability. The most probable value is somewhere in the range of one thousand. The actual intervals shown here were arbitrarily selected. Any appropriate intervals may be used.

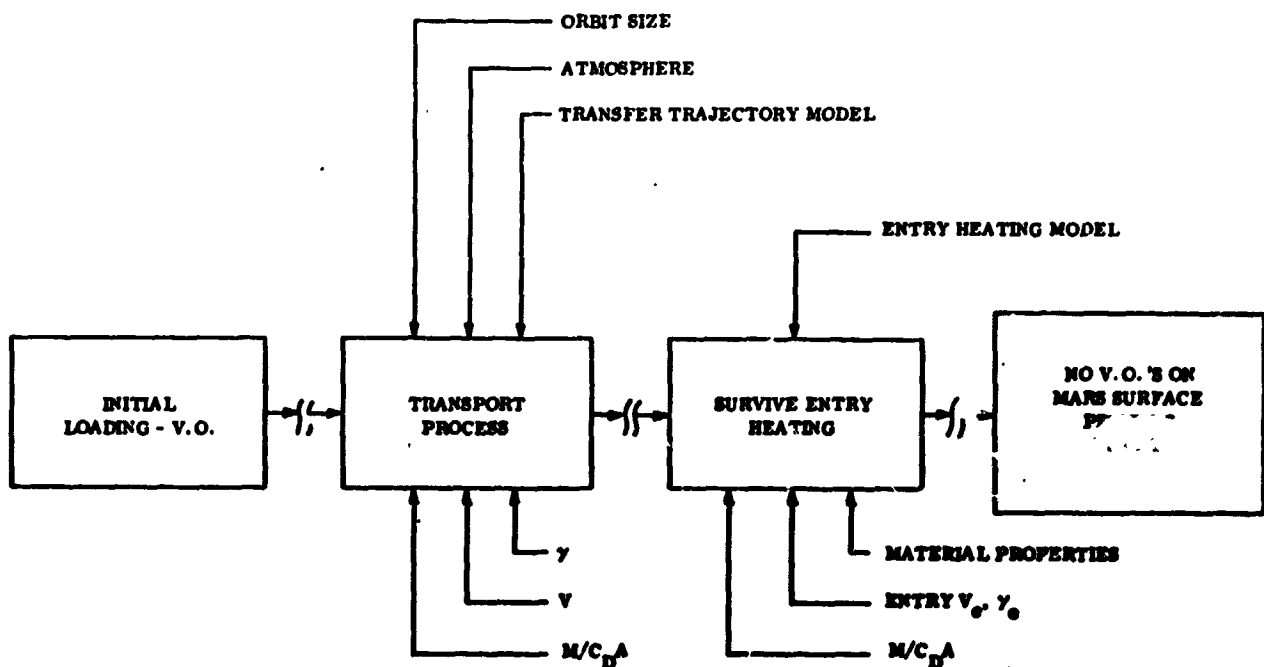


Figure 1-2. Computational Flow Diagram

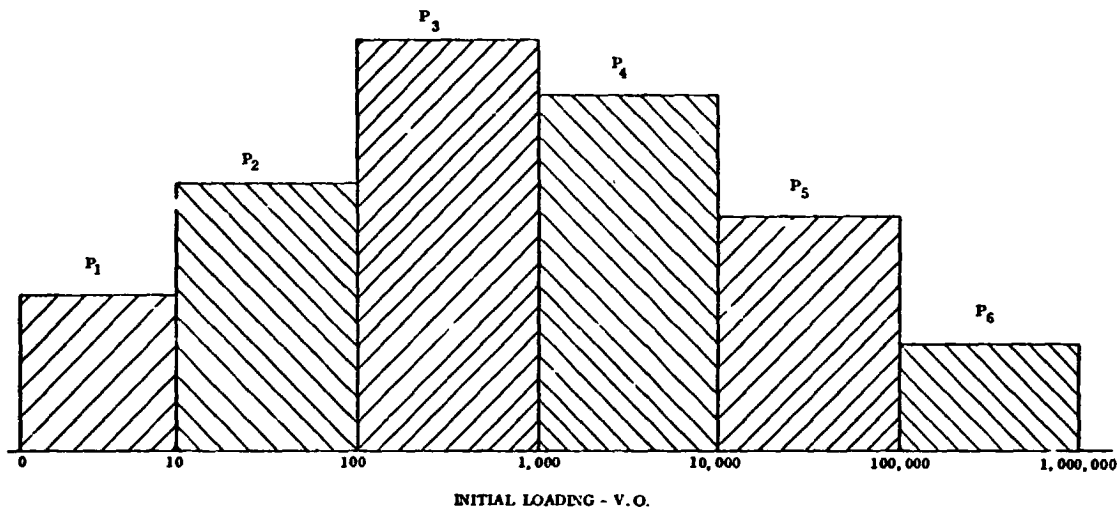


Figure 1-3. Input Data Format

Consider next the transport process. By this point in the problem, the manner in which particles leave the spacecraft has been considered. The trajectory the ejected particles take must be considered to determine if they can impact the surface of Mars by time T. An analysis must be made to determine if the particle will enter the atmosphere and eventually impact the surface of Mars. Some of the significant parameters associated with this analysis are the $\frac{M}{C_D A}$ of the particle, the velocity at which the particle leaves the spacecraft, and the angle at which it leaves the spacecraft. This analysis has been described in detail in document number VOY-C2-TR7.

Although the particle may take on a trajectory that will cause it to reach Mars, viable organisms carried by this particle may be killed enroute. As an example of one of the kill mechanisms, consider the atmosphere entry heating. The time-temperature history of the particle as it enters the Mars atmosphere and continues through the atmosphere to the surface of Mars must be considered to determine if the organisms will survive this kill mechanism. Parameters associated with this analysis are again the $\frac{M}{C_D A}$ of the particle, the velocity and angle at which the particle enters the atmosphere of Mars, and the material properties of the particle, such as emissivity. This analysis is discussed in Section 5 of this report.

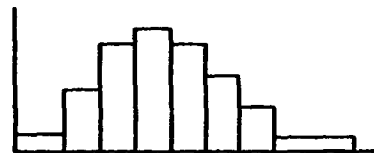
Figure 1-4 illustrates the type data that might be applicable to the description of the parameters associated with the transport process and the atmosphere entry heating. The three types of density functions that we encounter are a) the smooth continuous probability density function, b) a continuous probability density function having a finite number of intervals with the density uniform over each interval, and c) a discrete type probability density function which takes on only certain values, such as integers, each value having some probability of occurrence. Each of these probability density functions may be approximated by either of the other two. Parameters such as $\frac{M}{C_D A}$, velocity, angle, emissivity, etc., are normally described by continuous type density functions. Such parameters as number of viable organisms, however, are best described by discrete type density functions. If it is sufficient to describe the number as lying within some range, however, then the second type of density function may be used to describe the number of viable organisms.

1.1.3 OUTPUT

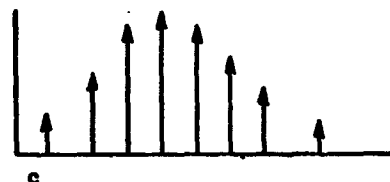
Consider now the number of viable organisms that reach the Mars atmosphere or surface and survive before time T, which is the output of the contamination analysis for a given source. Figure 1-5 illustrates the type density function that might represent this output. A discrete type density function is used to represent the number of viable organisms in the region near zero. This allows one to look at the probability of getting one or more organisms, two or more organisms, three or more organisms, etc. For the larger numbers it may be sufficient merely to state the probability that the number lies within some range or ranges, such as between five and one hundred, or between one hundred and one million. Based on the hypothetical information in Figure 1-5, the probability that one or more viable organisms



a



b



c

Figure 1-4. Alternate Data Formats

will reach Mars surface and continue to survive before time T is 10^{-3} . This is arrived at by either adding up the probabilities for 1, 2, 3, 4, 5, and on up, or by subtracting the probability of zero (which is 0.999) from 1. This hypothetical density function also shows that the probability of four or more viable organisms reaching Mars surface and continuing to survive before time T is 10^{-4} . If, for example, the criteria for contamination of Mars was one or more organisms reaching Mars and the constraint is that the probability of contaminating Mars must not exceed 10^{-4} , then the constraint would not have been met based on the information shown in the figure. If, on the other hand, as many as three viable organisms reaching Mars could be tolerated without representing contamination, then the constraint of 10^{-4} would have been met. If it were to cost a great deal more to meet the constraint of 10^{-4} using the criteria of one or more viable organisms representing contamination, then some reevaluation of the problem may be in order.

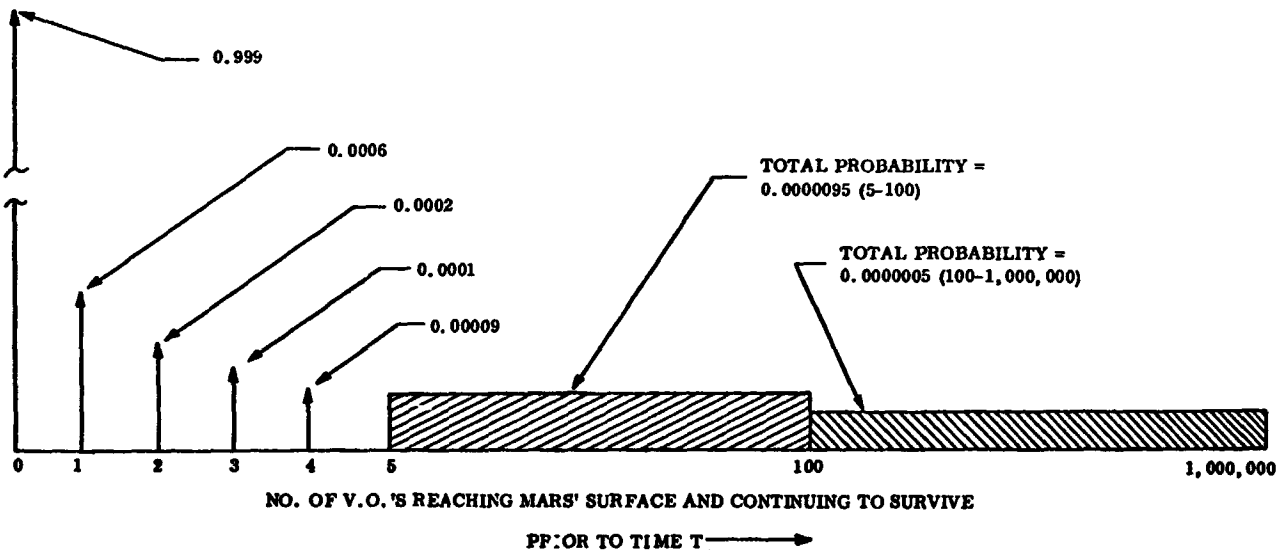


Figure 1-5. Typical Output Format

1.2 COMPUTATIONAL PROBLEMS

Primarily, there are three distinct kinds of computational problems associated with the determination of the probability of contaminating Mars before time T. The first is that of

operating on an initial probability density function by a set of conditional probability density functions and thus generating a marginal probability density function as the output. The second is that of summing probability density functions. The third is that of generating the probability density function on a random variable that has been defined as some function of one or more other random variables, each having its own a priori probability density function.

1.2.1 CONDITIONAL PROBABILITY DENSITY FUNCTIONS

This computation arises in situations such as:

- a. Given a range of sizes of particles (S) (probability density function on size), what are the number of viable organisms (O) carried by these particles? This computation is represented graphically by Figure 1-6.

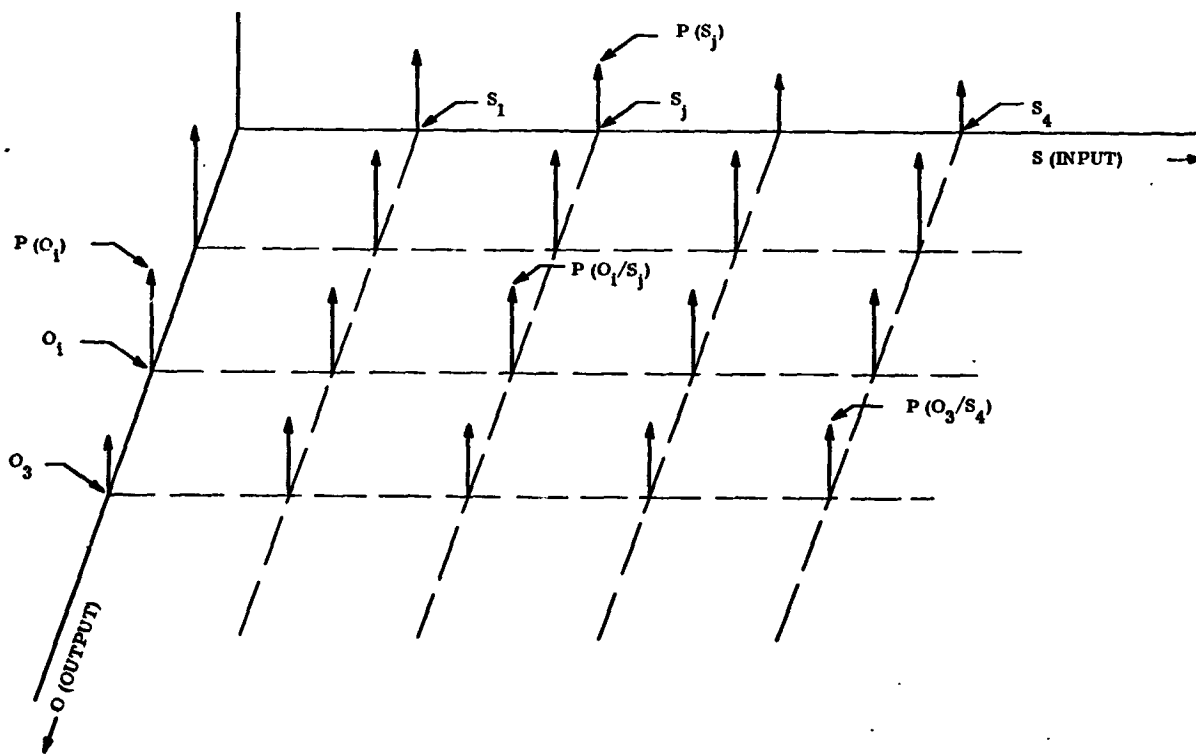


Figure 1-6. Conditional Probability Density Functions

The output (or marginal) distribution is computed as follows:

$$p(O_1) = p(O_1 | S_1) \cdot p(S_1) + p(O_1 | S_2) \cdot p(S_2) + p(O_1 | S_3) \cdot p(S_3) + p(O_1 | S_4) \cdot p(S_4)$$

$$p(O_2) = p(O_2 | S_1) \cdot p(S_1) + p(O_2 | S_2) \cdot p(S_2) + p(O_2 | S_3) \cdot p(S_3) + p(O_2 | S_4) \cdot p(S_4)$$

$$p(O_3) = p(O_3 | S_1) \cdot p(S_1) + p(O_3 | S_2) \cdot p(S_2) + p(O_3 | S_3) \cdot p(S_3) + p(O_3 | S_4) \cdot p(S_4)$$

In general notation, the above can be written in one statement as follows:

$$p(O_i) = \sum_{j=1}^{j=4} p(O_i | S_j) \cdot p(S_j) \quad ; \quad i = 1, 2, 3$$

- b. Given a range of number of viable organisms that are subjected to a certain kill mechanism (X) (prob. density function on number), what are the numbers of viable organisms (Y) that survive the kill mechanism? This computation is represented graphically by Figure 1-7.

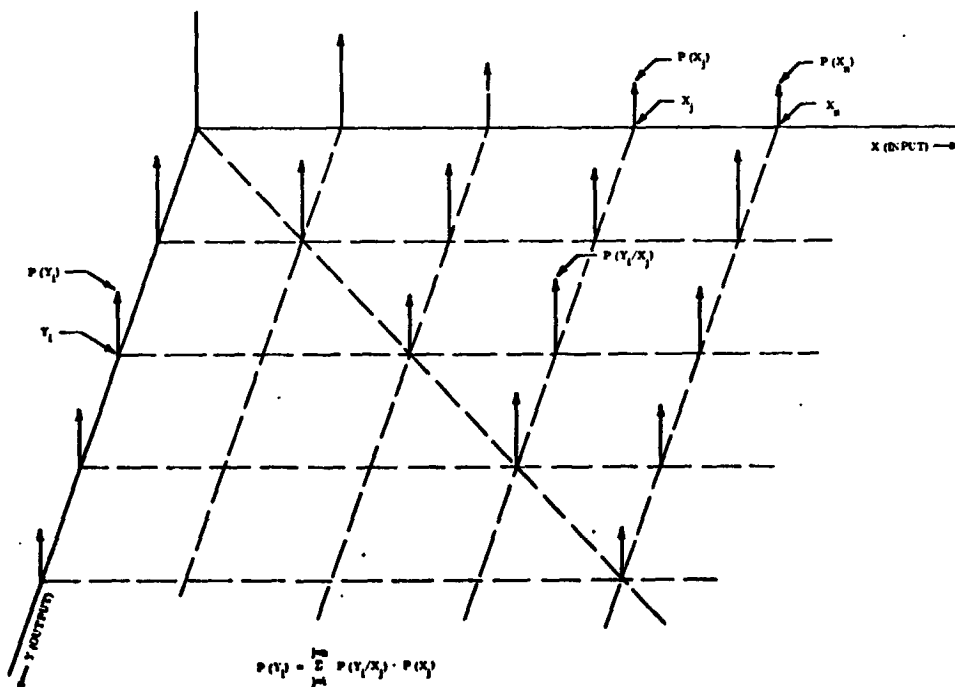


Figure 1-7. Conditional Probability Density Functions

1.2.2 SUMMATION OF PROBABILITY DENSITY FUNCTIONS

This computation arises, primarily, in arriving at the total number of viable organisms that will reach the surface of Mars as a result of all possible contamination sources.

Conceptually, the computation consists of adding all possible combinations of numbers from the several probability density functions, computing the probability of each combination, establishing groups of common sums of numbers, and then computing the probability of each group of common sums. A very simplified illustration of this is given in Figure 1-8 and Tables 1 and 2. Two discrete probability density functions are considered for this purpose.

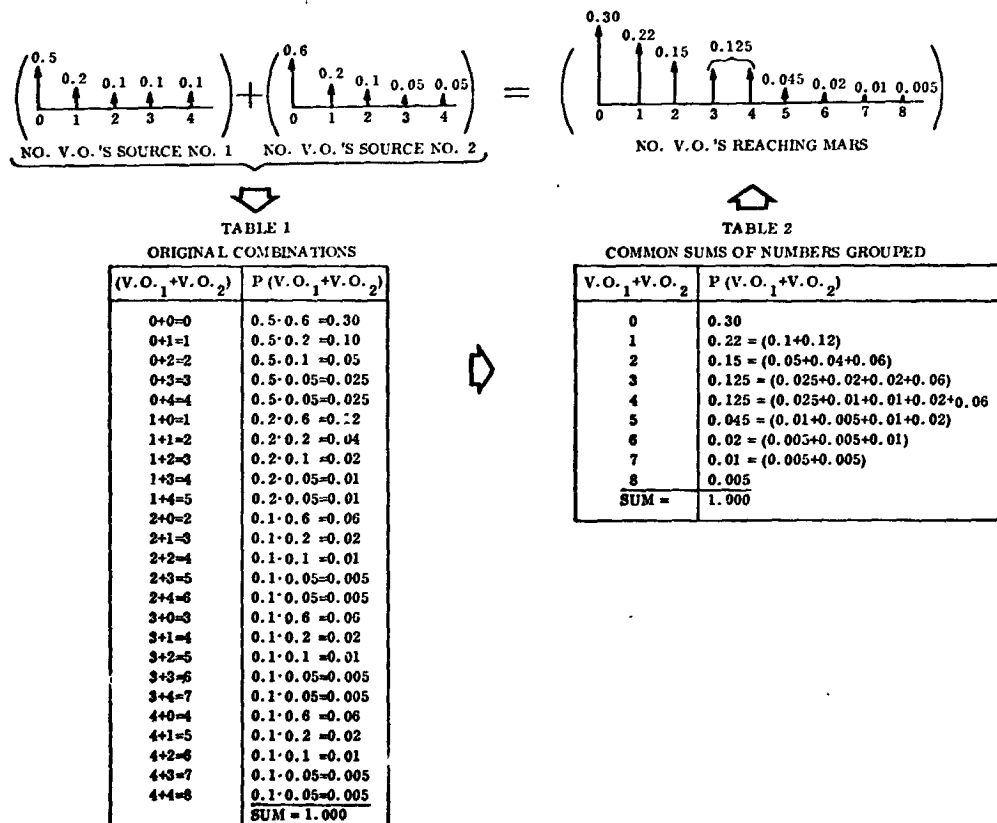


Figure 1-8. Summation of Probability Density Functions

1.2.3 COMBINATION OF PROBABILITY DENSITY FUNCTIONS ACCORDING TO SPECIFIED EQUATION

This type computation arises in the interplanetary trajectory analysis, orbit mechanics analysis, atmosphere entry heating analysis, etc. Conceptually, the computation is done in the same manner as for the summation of random variables. Most equations of interest, obviously, are much more complex than a simple summation, but the technique is essentially the same.

1.3 IMPLEMENTATION

There are four basic ways of implementing the computational process described in the previous section.

1.3.1 CLOSED FORM SOLUTION

This approach requires that the probability density functions be described in closed mathematical form. The data from which these density functions are derived rarely lend themselves to a closed form mathematical description. However, some density functions may be described in this form either directly or through curve fitting.

Even when all of the density functions of interest are known in closed mathematical form, it is generally not possible to perform the necessary mathematical integrations in order to arrive at a closed form solution of the problem of interest. This difficulty becomes greatly magnified when the density functions must be combined according to some complex equation.

1.3.2 MONTE CARLO SIMULATION

Conceptually, this is a very simple technique in terms of formulating the problem. It consists, essentially, of randomly selecting a value from each density function, operating on the set of values in the appropriate manner (i.e., summation, multiplication, division, or by some complex equation) and then repeating this process a sufficient number of times until the true density solution has been closely simulated.

The shortcoming of this approach is that, when a large number of random variables are involved, the number of samples required by the Monte Carlo approach to simulate the true answer is extremely large, thus requiring a large amount of high speed computer time. A further difficulty is that there is no technique available to determine in advance just how large the sample must be in order to approach the true answer with a given level of accuracy.

1.3.3 NUMERICAL APPROACH (DISCRETE VALUES)

This technique calls for combining all possible values of the parameters and computing the probability of each combination so that an output value is generated with an associated

probability. After all combinations have been generated, the output values are then grouped into common groupings and the probability of each group is computed, thus describing the probability density function for the output. The total number of combinations can be extremely large if there are a large number of parameters to be considered simultaneously. Quite often the approach becomes impractical for this reason.

1.3.4 NUMERICAL APPROACH (INTERVAL CONCEPT)

This approach is also a straightforward technique of using discrete approximations of continuous functions, considering all possible combination of values, computing the probability of each combination of values, operating on the set of values appropriately to generate the output value, grouping similar output values and then computing the probability of each group. Each density function is divided into intervals so that the combination of values referred to above are combinations of intervals of values and not combinations of discrete values. Furthermore, the density functions are combined pairwise in such a manner as to reduce the total number of combinations under consideration.

To illustrate the interval approach a transfer function ($V = W \cdot X + Y/Z$), such as shown in Figure 1-9, is assumed. W, X, Y and Z each have probability density functions as shown. The density functions are truncated at lower and upper values and the parameters W, X, Y, Z may take on any values within the range of their respective lower and upper values. The figures should be interpreted as follows: Looking first at the probability density function on W---"The probability that the value lies between 2 and 3 is 0.1; between 3 and 4 is 0.7; and between 4 and 5 is 0.2.

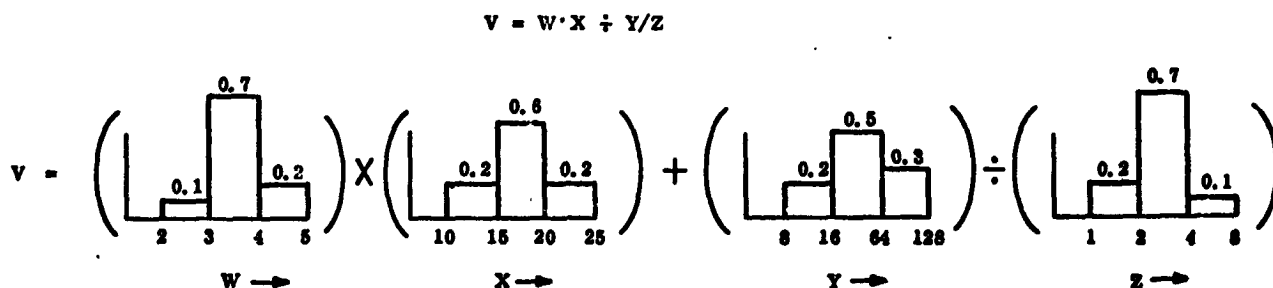


Figure 1-9. Intervals of Variables Problem

The probability density functions on X, Y, and Z are interpreted in the same way as those for W. However, there is one difference. The widths of the intervals on W and X are not uniform. As stated earlier, any appropriate widths of intervals may be used - they need not be uniform.

Figure 1-10 illustrates the combination of parameters by considering the first part of the transfer function $V = W \cdot X + Y/Z$ ---; that is, $W \cdot X$. Consider first the interval 2 - 3 on W and the interval 10 - 15 on X. All values of $W \cdot X$ resulting from these intervals will lie in a new interval having as a lower value $2 \cdot 10 = 20$, and an upper value $3 \cdot 15 = 45$. The probability associated with this new interval is $0.1 \cdot 0.2 = 0.02$. This is simply the probability that the value of W lies between 2 and 3 and the value of X lies between 10 and 15. The output of this combination of intervals is shown in Row 1 of the table. Columns 1 and 2 show the lower and upper limits of the new interval generated by combining the first interval of W with the first interval of X. Column 3 shows the probability that the new value $W \cdot X$ lies in the new interval. Columns 4, 5 and 6 are representative of the intervals into which the outputs of the transfer of $W \cdot X$ are to be grouped. The probability shown in Column 3 that is associated with the interval indicated by Columns 1 and 2 is appropriately prorated into Columns 4, 5 and 6. This process is repeated for each combination of intervals. Because W has three intervals and X has three intervals, there are nine combinations of intervals to be considered.

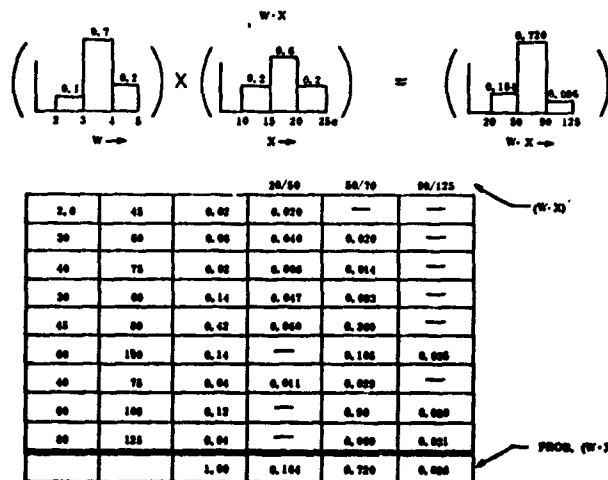


Figure 1-10. Intervals of Variables Solution

The intervals into which the outputs of the transfer of $W \cdot X$ are to be grouped have been arbitrarily designated as 20-50, 50-70 and 70-125.

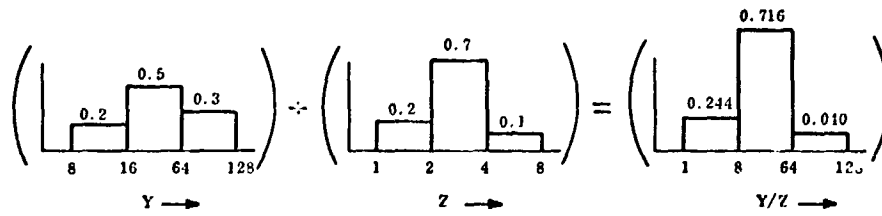
The new interval generated by the first combination of intervals is 20-45 with an associated probability of 0.02 (this is shown in the first three columns - Row 1 of the table). Because this interval is wholly contained by the first interval of the output density function, the entire 0.02 is put into the 20-50 interval.

Consider now the second combination of intervals (i. e. , - $W \rightarrow 2-3$ and $X \rightarrow 15-20$). The new interval generated by this combination will have as its lower limit $2 \cdot 15 = 30$ and as its upper limit $3 \cdot 20 = 60$. The probability associated with this interval 30-60 is $0.1 \cdot 0.6 = 0.06$ (this is shown in the first three columns - Row 2 of the table). This probability is prorated to the 20-50 interval by the ratio $30-50/30-60$, and to the 50-70 interval by the ratio $50-60/30-60$ (i. e. , $\frac{20}{30} \times 0.06 = 0.04$ is put into the 20-50 output interval and $\frac{10}{30} \times 0.06 = 0.02$ is put into the 50-70 output interval). The remainder of the nine combinations is done exactly in the same manner. The probabilities in each of the output intervals are now totaled thus defining the output probability density function. That is, the probability that the value of the output, $W \cdot X$, lies in the interval.

• 20-50 is 0.184 • 50-70 is 0.720 • 70-125 is 0.096

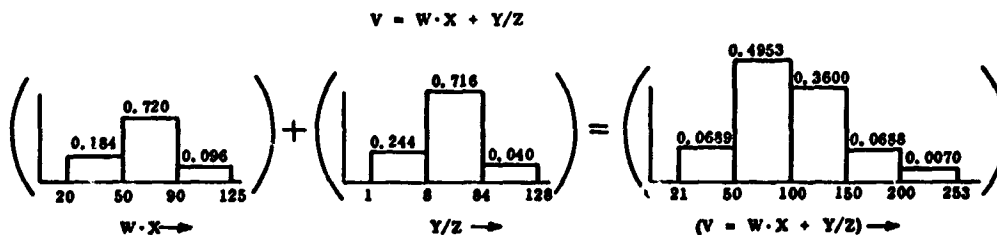
The solution generated by this technique approaches the exact solution as we consider more and more intervals on the input parameters (i. e. , as the width of the intervals approach zero). This allows the analyst to test for convergence to "sufficient" accuracy as the number of intervals considered are increased.

The rest of the computations for generating the output probability density function on V , when V has been defined by $V = W \cdot X + Y/Z$ (shown in Figures 1-11 and 1-12), are accomplished in basically the same manner as we have shown here. Figure 1-13 graphically represents the total process. First operate on W and X according to the transfer function to produce the output $W \cdot X$. Then operate on Y and Z according to the transfer function to produce Y/Z . Finally, operate on $(W \cdot X)$ and (Y/Z) according to the transfer function to produce $(V = W \cdot X + Y/Z)$.



		1/8	8/64	64/128	(Y/Z)
4	15	0.04	0.013	0.027	-
2	8	0.14	0.140	-	-
1	4	0.02	0.020	-	-
8	64	0.10	-	0.100	-
4	32	0.35	0.050	0.300	-
2	16	0.05	0.021	0.029	-
32	128	0.06	-	0.020	0.040
16	64	0.21	-	0.210	-
8	32	0.03	-	0.030	-
		1.00	0.244	0.716	0.040

Figure 1-11. Intervals of Variables Solution



		21/50	50/100	100/150	100/200	200/253	(V = W·X + Y/Z)
21	50	0.0449	0.0352	0.0097	-	-	-
28	114	0.1317	0.0337	0.0766	0.0214	-	-
84	178	0.0074	-	0.0013	0.0039	0.0022	-
51	98	0.1757	-	0.1757	-	-	-
58	154	0.5155	-	0.2255	0.2685	0.0215	-
114	218	0.0288	-	-	0.0190	0.0138	0.0056
91	133	0.0234	-	0.0050	0.0184	-	-
98	189	0.0688	-	0.0015	0.0078	0.0226	-
184	253	0.0038	-	-	-	0.0018	0.0000
		1.0000	0.0688	0.4852	0.2600	0.0055	0.0070

Figure 1-12. Intervals of Variables Solution

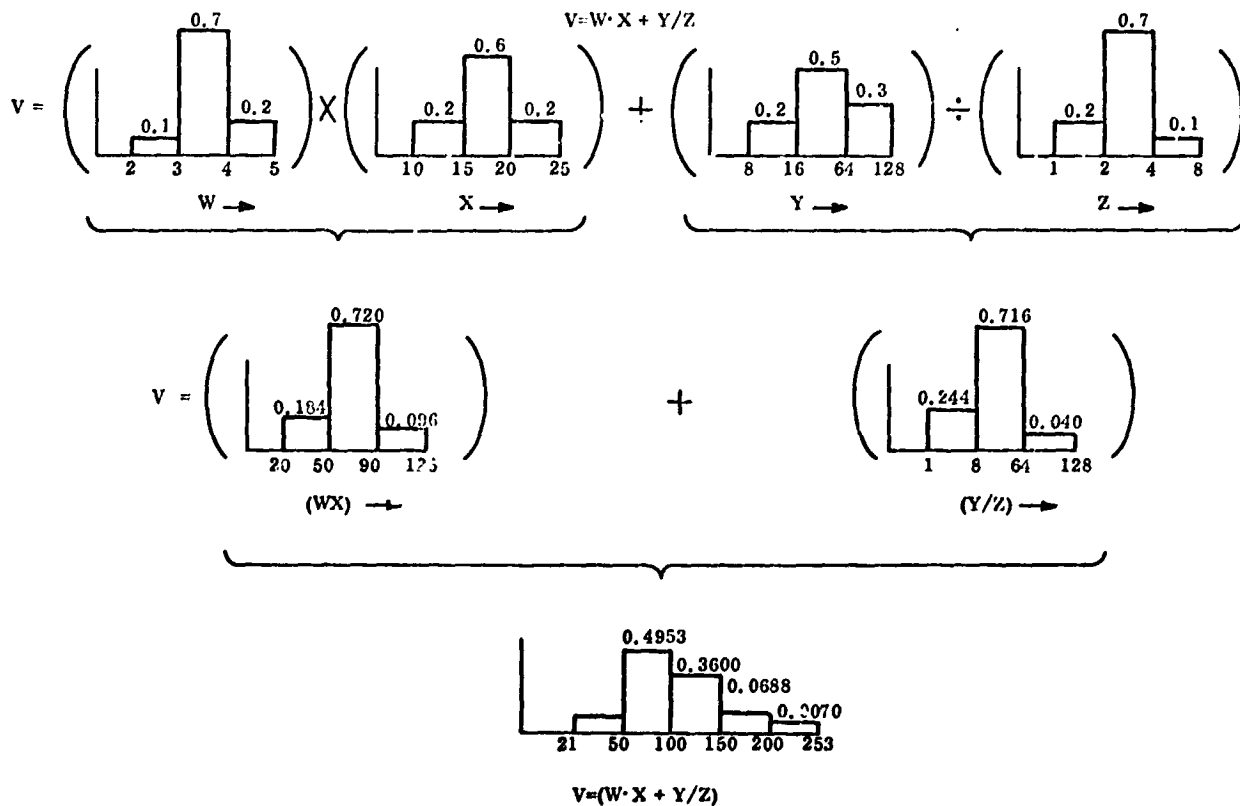


Figure 1-13. Intervals of Variables Solution

Not that the input and output density functions may have any number of intervals - they may vary in number from density function to density function. In the case of some parameters, a large number of intervals may be appropriate while in the case of other parameters a small number of intervals may be adequate.

The following is a brief summary of the computation methods discussed above.

- a. An analytical solution - that is, arriving at the output density function as a mathematical expression in closed form - is generally not practical. The input functions usually cannot be expressed as closed form mathematical expressions. Even when it is possible to do so, the solution for arriving at the output (i.e., the multiple integrations that must be performed) generally becomes intractable.
- b. A numerical solution using a Monte Carlo simulation technique is at best questionable. Because this is basically a sampling technique, the question is - how many samples must be taken in order to "adequately" simulate the output? A good method for arriving at an answer to this question is not available. Industry's experience has indicated that, in general, a very large number of samples are required, thus requiring a lot of computer time. The problem gets especially severe when there are a large number of input random variables to be considered.

- c. A numerical approach whereby an input probability density function is represented by a set of discrete values each with an associated probability of occurrence---all possible combinations of values from the parameters are generated with its associated probability. The concept is very simple but there may also be time of computation drawbacks associated with this technique. For example, let us consider the problem of arriving at the total number of viable organisms reaching the surface of Mars. Assuming that we have done this for each possible source of contamination and further assuming that there are 100 sources, we must now sum up over all 100 sources. If we were to consider all possible combinations of sums, we would have a very large number to compute. If each density function were described by only 3 discrete values there would be 3^{100} sums to compute. (This is approximately equal to 10^{50} .) If each calculation were to take 10^{-5} seconds on a computer, the total number of seconds of computer time required would be $10^{50}/10^5 \approx 10^{45}$ seconds or approximately 10^{38} years of continuous computer operation. If each density function was represented by 100 discrete values, the total number of sums to compute would be 100^{100} or 10^{200} . It is immediately obvious that this is not the method to use.
- d. If the operation of summing was done pairwise, the total number of sums to be computed would be drastically reduced. Using the same example as before (i.e., 100 density function each having 3 discrete values) the technique is to:
- sum the first two density functions thus generating 3^2 or 9 sums
 - group these values and approximate the resulting density function by 3 discrete values
 - continue this process until the 100th density function has been added thus defining the final total.

The total number of sums that would have been generated is $(100-1) \cdot 3^2$ or approximately 900. It is true that three points are not very representative of most probability density functions. Therefore, consider 100 points on each density function. The pairwise approach would yield a total of $(100-1) 100^2$ or approximately 10^6 sums. If, as before, each calculation took 10^{-5} sec of computer time, the time required would be $10^6/10^5 = 10$ sec. It is obvious then, that this is well within practical limits.

Now if the "interval concept" is considered which considers density functions pairwise, we see that the number of calculations are even further reduced. For example, if we consider 20 intervals of values instead of 100 discrete values, we reduce the last number of calculations

by a factor of $\left(\frac{20}{100}\right)^2 = \frac{1}{25}$. It should be remembered that the interval approach allows you to constantly know the upper and lower bounds of the true value of the output. This feature is required so that the analyst will have confidence in the accuracy of his solution, given a set of input data.

1.4 COMPUTER PROGRAM DEVELOPMENT

To implement the calculations that have been described, a set of programs have been developed that may be used on the remote access computer system. (also called the Desk Side Computer System). The computational difficulties are somewhat minimized when one considers that high speed computers are available for the task. The problem is further minimized when one considers that desk side computer systems are available with the capability of having many different programs in storage for immediate callup. This type of system enables the analyst to work in his own office (or work area). An advantage of this is that he has the appropriate information and material immediately (and conveniently) available to him and he can work at the computer as problems arise rather than storing up problems for one big computer run.

A group of subprograms, such as described in Section 2 of this report, enables the analyst to work on the total problem in pieces. This allows him to (1) become completely familiar with the inner workings of the program and (2) give him the ability to make sensitivity studies on parameters within confined areas of the total problem. A further advantage of having a group of subprograms, rather than one large integrated program, is that the analyst may wish to change the program in certain areas. It is very difficult to do this when the program has been developed as one integrated program.

SECTION 2
MECHANIZATION

By
T. Green

TABLE OF CONTENTS

Section	Page
2 MECHANIZATION	2-1
2.1 Basic Numerical Process	2-1
2.2 Binomial Probabilities	2-4
2.3 Basic Combinatorial Model	2-7
2.4 Addition of Random Variables	2-13
2.5 Dependent Probabilities	2-20
2.6 Marginal Probability Calculation	2-22
2.7 Periapsis Distribution Determination	2-25
2.8 Mars Orbit Time and M/C_{DA} Distribution	2-30
2.9 Heliocentric Transfer Case	2-34
2.10 Entry Survival Probability	2-44
2.11 M/C_{DA} Survival and Scale Probabilities	2-47
2.12 Scale Probabilities	2-49
2.13 General Combining of Random Variables	2-57
2.14 References for Section 2	2-59
2.15 Acknowledgements	2-59

SECTION 2
THE DESK SIDE VERSION OF THE
VOYAGER PLANETARY QUARANTINE MATH MODEL

2.1 BASIC NUMERICAL PROCESS

The math model consists of a package of programs which help to provide insight into the various sensitivities of the effect of parameter values on random variables.

This section describes the key concepts involved in the development of the computer programs and contains a brief explanation of how to use each program.

The basic numerical process incorporated in many of the programs is the method of loading an interval probability onto a selected grid pattern. This technique is used for both discrete and continuous or interval probabilities. In the following runs the probability of 1. at 0., 0. to 1., 3., 0. to 3., 0. to 4., .5 to 1.2, 1. to 3.5, and 1.5 to 3, respectively, were loaded into the basic grid pattern:

0., 0., 1., 1., 2., 2., 3., 3., 4., 4.

Repeated numbers indicate that a discrete probability could be assigned. The logic is as follows:

If a discrete probability is given, it will either be loaded directly onto a discrete grid point or added into the appropriate interval. For an interval, no probability will be added onto discrete points but will be apportioned proportionately onto the grid pattern.

PROGRAM TO LOAD PROBABILITIES

NUMBER, OUTPUT GRID VALUES:=10,0,0,1,1,2,2,3,3,4,4.

INTERVAL START, END, PROBABILITY:=0,0,1.

RESULTING PROBABILITIES

1.000000E+00	0.000000E-01	0.000000E-01	0.000000E-01	0.000000E-01
0.000000E-01	0.000000E-01	0.000000E-01	0.000000E-01	0.000000E-01

INTERVAL START, END, PROBABILITY:=0,1,1.

RESULTING PROBABILITIES

0.000000E-01	1.000000E+00	0.000000E-01	0.000000E-01	0.000000E-01
0.000000E-01	0.000000E-01	0.000000E-01	0.000000E-01	0.000000E-01

INTERVAL START, END, PROBABILITY:=3,3,1.

RESULTING PROBABILITIES

0.000000E-01	0.000000E-01	0.000000E-01	0.000000E-01	0.000000E-01
0.000000E-01	1.000000E+00	0.000000E-01	0.000000E-01	0.000000E-01

INTERVAL START, END, PROBABILITY:=0,3,1.

RESULTING PROBABILITIES

0.000000E-01	3.333333E-01	0.000000E-01	3.333333E-01	0.000000E-01
3.333333E-01	0.000000E-01	0.000000E-01	0.000000E-01	0.000000E-01

INTERVAL START, END, PROBABILITY:=0.,4.,1.

RESULTING PROBABILITIES

0.000000E-01	2.500000E-01	0.000000E-01	2.500000E-01	0.000000E-01
2.500000E-01	0.000000E-01	2.500000E-01	0.000000E-01	

INTERVAL START, END, PROBABILITY:=.5,1.2,1.

RESULTING PROBABILITIES

0.000000E-01	7.142857E-01	0.000000E-01	2.857143E-01	0.000000E-01
0.000000E-01	0.000000E-01	0.000000E-01	0.000000E-01	

INTERVAL START, END, PROBABILITY:=1.,3.5,1.

RESULTING PROBABILITIES

0.000000E-01	0.000000E-01	0.000000E-01	4.000000E-01	0.000000E-01
4.000000E-01	0.000000E-01	2.000000E-01	0.000000E-01	

INTERVAL START, END, PROBABILITY:=1.5,3.,1.

RESULTING PROBABILITIES

0.000000E-01	0.000000E-01	0.000000E-01	3.333333E-01	0.000000E-01
6.666667E-01	0.000000E-01	0.000000E-01	0.000000E-01	

2.2 BINOMIAL PROBABILITIES

A fundamental distribution which appears to be basic to much of our study is the binomial. If the probability that an organism survives an event is θ , then in repeated trials (n), the probability of survival X times is

$$f(x) = \binom{n}{x} \theta^x (1-\theta)^{n-x}$$

The mean of this distribution is $n\theta$, which gives an indication of its shape. For smaller θ the distribution piles up about $x = 0$ and becomes increasingly skewed to the right.

The numerical evaluation of this distribution becomes difficult if the above expression is used directly.

Since

$$\frac{f(x+1)}{f(x)} = \left(\frac{n-x}{x+1} \right) \left(\frac{\theta}{1-\theta} \right) = R(x)$$

The recursion $f(x+1) = R(x) f(x)$

where $f(0) = (1-\theta)^n$ allows us to evaluate this distribution without the calculation of factorials or other combinatorial formulas.

A program called "BINOM" performs the above.

For large n , the recursion becomes less attractive for obvious reasons. In this case, if

$$\frac{1}{n+1} < \theta$$

The binomial can be approximated by the normal in a 3σ range about the mean. Denoting the cumulative values of the normal by $\Phi(x)$, the approximation is

$$\Pr \left\{ a \leq x \leq b \right\} \triangleq \Phi \left(\frac{b + \frac{1}{2} - \mu}{\sigma} \right) - \Phi \left(\frac{a + \frac{1}{2} - \mu}{\sigma} \right)$$

$$\mu = n\theta, \quad \sigma = \sqrt{n\theta(1-\theta)}$$

A program called "BINOMX" performs the above calculations by utilizing a technique developed and described in Reference 3.

It is recommended that "BINOMX" be used for $n > 100$.

The two following sets of examples illustrate the output of BINOM and BINOMX.

In BINOMX, the equivalent upper and lower bounds for the given first and last values are in terms of the standard normal distribution.

Note that essentially all (99.99 percent approximately) of the probability lies between ± 3 .

BINOMIAL PROBABILITIES		READ PROBABILITY AND NUMBER:=.1.10	
READ PROBABILITY AND NUMBER:=.1.0		VALUE	PROBABILITY
VALUE	PROBABILITY	0	3.486784E-01
0	1.000000E+00	1	3.874205E-01
SUM=	1.0000000	2	1.937102E-01
READ PROBABILITY AND NUMBER:=.1.3		3	5.739563E-02
VALUE	PROBABILITY	4	1.116026E-02
0	7.290000E-01	5	1.488035E-03
1	2.430000E-01	6	1.377810E-04
2	2.700000E-02	7	8.748000E-06
3	1.000000E-03	8	3.645000E-07
SUM=	1.0000000	9	9.000000E-09
READ PROBABILITY AND NUMBER:=.1.7		10	1.000000E-10
VALUE	PROBABILITY	SUM=	1.0000000
0	4.782969E-01		
1	3.720087E-01		
2	1.240029E-01		
3	2.296350E-02		
4	2.551500E-03		
5	1.701000E-04		
6	6.300000E-06		
7	1.000000E-07		
SUM=	1.0000000		

NORMAL APPROXIMATION TO BINOMIAL

READ FIRST, LAST VALUE, NUMBER, PROBABILITY:=10., 50., 550., .1

STANDARD NORMAL LIMITS***** -0.646709E+01 -0.639602E+00
PROBABILITY= 0.261216E+00

READ FIRST, LAST VALUE, NUMBER, PROBABILITY:=10., 100., 550., .1

STANDARD NORMAL LIMITS***** -0.646709E+01 0.646709E+01
PROBABILITY= 0.100000E+01

READ FIRST, LAST VALUE, NUMBER, PROBABILITY:=20., 30., 550., .1

STANDARD NORMAL LIMITS***** -0.504575E+01 -0.348228E+01
PROBABILITY= 0.248358E-03

READ FIRST, LAST VALUE, NUMBER, PROBABILITY:=20., 25., 100., .5

STANDARD NORMAL LIMITS***** -0.610000E+01 -0.490000E+01
PROBABILITY= 0.478653E-06

READ FIRST, LAST VALUE, NUMBER, PROBABILITY:=25., 75., 100., .5

STANDARD NORMAL LIMITS***** -0.510000E+01 0.510000E+01
PROBABILITY= 0.100000E+01

READ FIRST, LAST VALUE, NUMBER, PROBABILITY:=30., 60., 100., .5

STANDARD NORMAL LIMITS***** -0.410000E+01 0.210000E+01
PROBABILITY= 0.982115E+00

READ FIRST, LAST VALUE, NUMBER, PROBABILITY:=40., 60., 100., .5

STANDARD NORMAL LIMITS***** -0.210000E+01 0.210000E+01
PROBABILITY= 0.964271E+00

READ FIRST, LAST VALUE, NUMBER, PROBABILITY:=45., 55., 100., .5

STANDARD NORMAL LIMITS***** -0.110000E+01 0.110000E+01
PROBABILITY= 0.728668E+00

READ FIRST, LAST VALUE, NUMBER, PROBABILITY:=47., 53., 100., .5

STANDARD NORMAL LIMITS***** -0.700000E+00 0.700000E+00
PROBABILITY= 0.516073E+00

READ FIRST, LAST VALUE, NUMBER, PROBABILITY:=\$STOP
READY.

2.3 BASIC COMBINATORIAL MODEL

The basic module in the numerical math model is the two-segment program titled "Basic 1, Basic 2." The primary function of the program is to combine probabilities in the following form:

		EVENTS				
		1	2	3		M
SOURCES	ROW 1					
	ROW 2					
	ROW n					

Viable organisms can be thought of as being available from various sources about the spacecraft and environs. The number available conceivably could be a random variable and thus can be represented as a "row" probability distribution. Each event (represented by a column) can be thought of as altering the row (source) probability distribution. The simplest way of representing the effect of an event is to say that an organism has probability of surviving the event. If it can be hypothesized that each organism has the same probability, then the conditional distribution of survival is of a binomial nature. That is, given n organisms, the probability of x survivors is

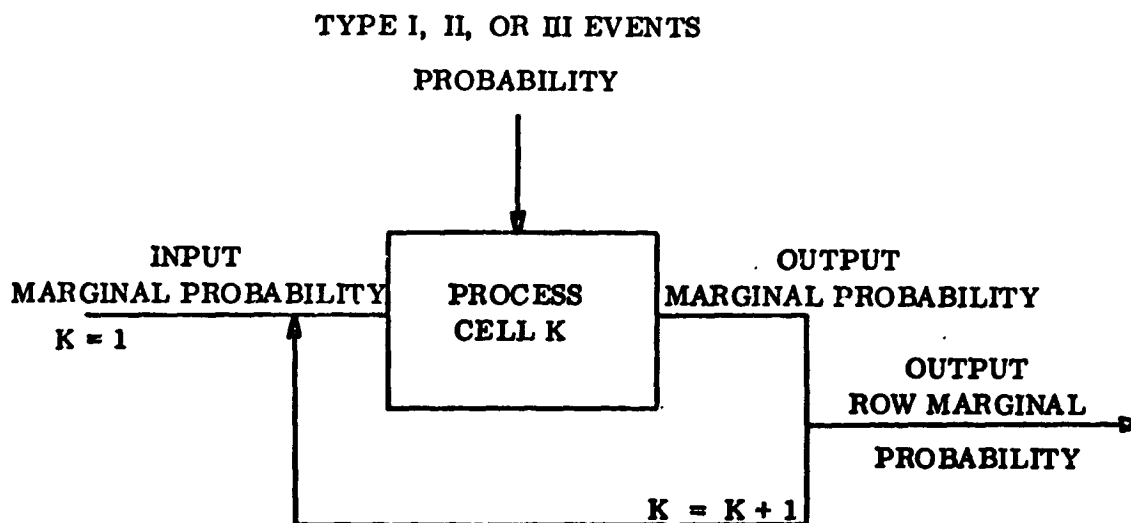
$$\text{pr}(x/n) = \binom{n}{x} \epsilon^x (1-\epsilon)^{n-x}$$

the resulting marginal probability

$$\text{pr}(x) = \sum_{n_1}^n \text{pr}(x) \binom{n_1}{x} \epsilon^x (1-\epsilon)^{n_1-x}$$

where N is the total number of organisms before the event. If the binomial hypothesis is not satisfactory any conditional distribution, can be used in place of $pr(x/n)$. It is primarily the function of this program to calculate each succeeding marginal probability as one courses through the events. As a row is completed, the resulting random variables can be added by another program called "BUGS" which is described elsewhere.

1. Read in basic GRID pattern.
2. Read in Row probability source distribution along with its own grid pattern.
3. Adjust source probability distribution to standard grid pattern.



1. Read in row probability end points.
2. Read in associated probabilities.
3. Read in the standard set of end points to be used during the course of row calculations.
4. Adjust the input probability distribution to the standard set of end points.

5. Read in code word:
 - 1: Binomial probabilities
 - 2: Conditional distributions at arbitrary nature
 - 3: Proportion, probability
 - 4: Scale the standard end points
 - 5: Return to choose another row distribution and thence to process another row.
6. Process according to code word and print the input and output marginal distributions.

The option is described below:

OPTION CODE

1. Conditional probabilities are computed by assuming a binomial condition is satisfactory. Appropriately, the probability that an organism survives an event is input initially.
2. The conditional distributions are provided one at a time starting at the second to lowest grid value. The distributions are provided in the standard interval concept.
3. The conditional distributions are provided by inputting two numbers for each distribution (θ , ξ). Each conditional distribution has the end points σ , $G(I) * \theta$, $G(I)$ with associated probabilities ξ and $1. - \xi$.
4. A simple scale change is provided here. If the scale factor is q , then the new standard grid pattern is $q * G(I)$ with the same probabilities.
5. This indicates that a new row probability is to be provided. Essentially this restarts the program.

The combinatorial technique is based on the fundamental concept of conditional and marginal probability. Reference 4 describes the approach.

The combinatorial procedure is at best a computer approximation and requires some explanation of the method. Figure 2-1 illustrates the procedure.

GP(I) represents the input probabilities.

G(I) represents the standard grid points.

$\bar{P}OUT(I)$ is the output (marginal) probability distribution.

PR(J)/G(I) describes the conditional probabilities.

The conditional distributions in option 1 are either generated by the basic binomial recursion formula or the normal approximation to the binomial (see Reference 33) if the given number is greater than 100.

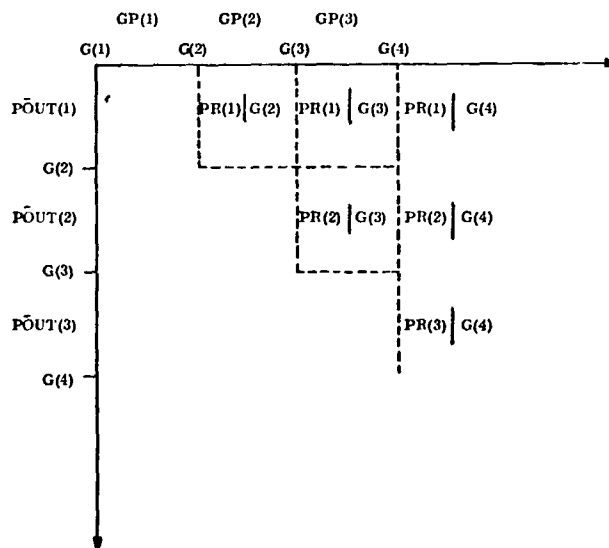
Options 2 and 3 require that the conditional distributions, be input via the interval concept.

The marginal probability distribution is calculated by

$$\bar{P}OUT(J) = \sum_{I=J+1}^{NST} GP(I-1) (PR(J)/G(I))$$

The probabilities GP (I) are associated with G(I+1). This appears to be the most conservative approach.

It is recommended that the first value for the standard grid be zero. This is consistent with the concept of the binomial hypothesis.



BASIC COMBINATORIAL MODEL

NUMBER, END POINTS:=4, 0., 2., 20., 1000.

ROW PROBABILITIES:=.6., .3., .1

NUMBER, STANDARD END POINTS:=10, 0., 1., 5., 10., 100., 1000., 2000., 3000., 4000., 5000.

READ IN CODE(1-5):=1

SIMPLE PROBABILITY:=.01

FIRST	LAST	INPUT	OUTPUT
0.000000E-01	1.000000E+00	3.000000E-01	8.614235E-01
1.000000E+00	5.000000E+00	3.500000E-01	5.175419E-02
5.000000E+00	1.000000E+01	4.333333E-02	4.089453E-02
1.000000E+01	1.000000E+02	1.745299E-01	4.593781E-02
1.000000E+02	1.000000E+03	9.183673E-02	0.000000E-01
1.000000E+03	2.000000E+03	0.000000E-01	0.000000E-01
2.000000E+03	3.000000E+03	0.000000E-01	0.000000E-01
3.000000E+03	4.000000E+03	0.000000E-01	0.000000E-01
4.000000E+03	5.000000E+03	0.000000E-01	0.000000E-01

READ IN CODE(1-5):=2

NUMBER, END POINTS FOR (1.000000E+00):=2, 0., 1.

CONDITIONAL PROBABILITIES:=1.

NUMBER, END POINTS FOR (5.000000E+00):=4, 0., 2., 3., 5.

CONDITIONAL PROBABILITIES:=.7., .2., .1

NUMBER, END POINTS FOR (1.000000E+01):=4, 0., 1., 5., 10.

CONDITIONAL PROBABILITIES:=.8., .2., .1

NUMBER, END POINTS FOR (1.000000E+02):=3, 0., 30., 100.

CONDITIONAL PROBABILITIES:=.9., .1

NUMBER, END POINTS FOR (1.000000E+03):=3, 0., 100., 1000.

CONDITIONAL PROBABILITIES:=.95., .05

NUMBER, END POINTS FOR (2.000000E+03):=3, 0., 1000., 2000.

CONDITIONAL PROBABILITIES:=.5., .5

NUMBER, END POINTS FOR (3.000000E+03):=3, 0., 2000., 3000.

CONDITIONAL PROBABILITIES:=1., .0.

NUMBER, END POINTS FOR (1.000000E+03):=2, 0., 4000.

CONDITIONAL PROBABILITIES:=1.

NUMBER, END POINTS FOR (5.000000E+03):=2, 0., 5000.

CONDITIONAL PROBABILITIES:=1.

FIRST	LAST	INPUT	OUTPUT
0.000000E-01	1.000000E+00	3.614235E-01	9.136232E-01
1.000000E+00	5.000000E+00	5.175419E-02	4.732967E-02
5.000000E+00	1.000000E+01	4.089453E-02	6.090672E-03
1.000000E+01	1.000000E+02	4.593781E-02	3.215647E-02
1.000000E+02	1.000000E+03	0.000000E-01	0.000000E-01
1.000000E+03	2.000000E+03	0.000000E-01	0.000000E-01
2.000000E+03	3.000000E+03	0.000000E-01	0.000000E-01
3.000000E+03	4.000000E+03	0.000000E-01	0.000000E-01
4.000000E+03	5.000000E+03	0.000000E-01	0.000000E-01

READ IN CODE(1-5):=3

PROPORTION, PROBABILITY FOR (1.00000E+00):=.3,.8
PROPORTION, PROBABILITY FOR (5.00000E+00):=.5,.65
PROPORTION, PROBABILITY FOR (1.00000E+01):=.8,.7
PROPORTION, PROBABILITY FOR (1.00000E+02):=.25,.1
PROPORTION, PROBABILITY FOR (1.00000E+03):=.1,.001
PROPORTION, PROBABILITY FOR (2.00000E+03):=1.,1.
PROPORTION, PROBABILITY FOR (3.00000E+03):=1.,1.
PROPORTION, PROBABILITY FOR (4.00000E+03):=1.,1.
PROPORTION, PROBABILITY FOR (5.00000E+03):=1.,1.

FIRST	LAST	INPUT	OUTPUT
0.00000E-01	1.00000E+00	9.136232E-01	9.266605E-01
1.00000E+00	5.00000E+00	4.732967E-02	3.795019E-02
5.00000E+00	1.00000E+01	6.890672E-03	4.519132E-03
1.00000E+01	1.00000E+02	3.215647E-02	3.087021E-02
1.00000E+02	1.00000E+03	0.000000E-01	0.000000E-01
1.00000E+03	2.00000E+03	0.000000E-01	0.000000E-01
2.00000E+03	3.00000E+03	0.000000E-01	0.000000E-01
3.00000E+03	4.00000E+03	0.000000E-01	0.000000E-01
4.00000E+03	5.00000E+03	0.000000E-01	0.000000E-01

READ IN CODE(1-5):=4

SCALE FACTOR:=20.

FIRST	LAST	INPUT	OUTPUT
0.00000E-01	1.00000E+00	9.266605E-01	4.633302E-02
1.00000E+00	5.00000E+00	3.795019E-02	1.853321E-01
5.00000E+00	1.00000E+01	4.519132E-03	2.316651E-01
1.00000E+01	1.00000E+02	3.087021E-02	5.012804E-01
1.00000E+02	1.00000E+03	0.000000E-01	1.823923E-02
1.00000E+03	2.00000E+03	0.000000E-01	1.715012E-02
2.00000E+03	3.00000E+03	0.000000E-01	0.000000E-01
3.00000E+03	4.00000E+03	0.000000E-01	0.000000E-01
4.00000E+03	5.00000E+03	0.000000E-01	0.000000E-01

READ IN CODE(1-5):=5

NUMBER, ROW END POINTS:=SSTOP
READY.

2.4 ADDITION OF RANDOM VARIABLES

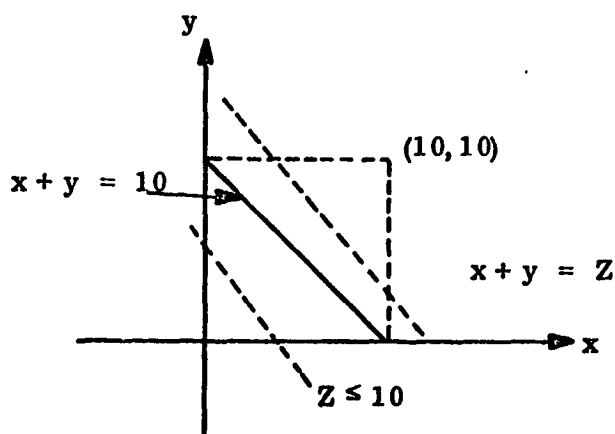
A common combinatorial problem is the computation of the distribution of the sum of two random variables. This occurs, for example, in the processing of combining row probabilities as described in Reference 4.

Consider the determination of the distribution of $Z = x + y$ where $0 \leq x \leq 10$, $0 \leq y \leq 10$, and it is assumed that the probability density functions of x and y are independent. (This is a basic assumption that will be made between rows or sources.)

The cumulative distribution

$$F(Z_0) = \Pr(Z \leq Z_0) = \iint_R f(x) g(y) dx dy$$

where R is the space in the x - y plane such that $Z \leq Z_0$.



The distribution can be explicitly written

$$\Pr(x+y \leq Z_0) = \int_0^{Z_0} \int_0^{Z_0-x} f(x) g(y) dy dx \quad (0 \leq Z_0 \leq 10)$$

$$\Pr (x+y \leq y \leq Z_0) = 1 - \int_{Z_0-10}^{10} \int_{Z_0-x}^{10} f(x) g(y) dy dx \quad (10 \leq Z_0 \leq 20)$$

and zero elsewhere.

For example, let $g(y) = \frac{1}{10}$; $f(x) = \frac{1}{10}$ for $0 \leq x \leq 10$ and $0 \leq y \leq 10$.

Then the resulting probabilities from $(0 \rightarrow 20)$ could be tabulated as follows:

<u>Interval</u>	<u>Probability</u>
0 → 2	0.02
2 → 4	0.06
4 → 6	0.10
6 → 8	0.14
8 → 10	0.18
10 → 12	0.18
12 → 14	0.14
14 → 16	0.10
16 → 18	0.06
18 → 20	0.02

where $\Pr (Z_1 \leq Z_0 \leq Z_2) = \frac{Z_2^2 - Z_1^2}{200} \quad (0 \leq Z_0 < 10)$

$$\Pr (Z_1 \leq Z_0 \leq Z_2) = \left(\frac{-Z_2^2}{200} + \frac{Z_0 Z_2}{100} - 1 \right) - \left(\frac{-Z_1^2}{200} + \frac{20Z_1}{100} - 1 \right)$$

$$(10 \leq Z_0 < 20)$$

If the probabilities $f(x)$ and $g(x)$ where given in more complex form, say as a mixed function defined as constant over prescribed intervals, it is more advantageous from a computer standpoint to develop an algorithm to "lump" probabilities assigned over prescribed intervals into the standard grid pattern.

The above problem was checked by using the intervals given above.

The following illustrates the results of a probability "adder."

The first number is a sum for the resulting probabilities and the following are the probabilities of Z in the intervals 0 → 2, 2 → 4, etc.

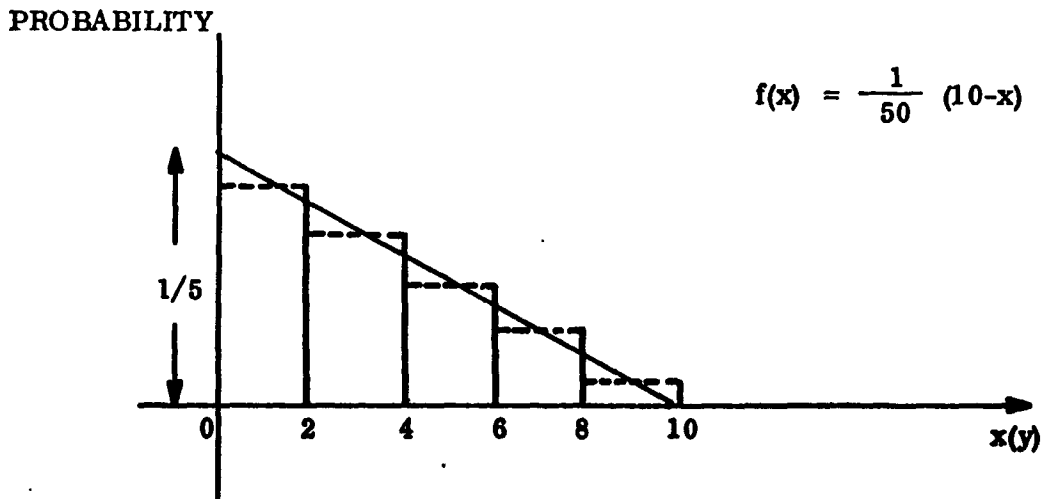
```

LOAD LIMITS 06345 15505
:=6,0.,2.,4.,6.,8.,10.
:=6,0.,2.,4.,6.,8.,10.
:=11,0.,2.,4.,6.,8.,10.,12.,14.,16.,18.,20.
:=.2,.2,.2,.2,.2
:=.2,.2,.2,.2,.2
      SUM
1.000000E+00  2.000000E-02  6.000000E-02  1.000000E-01  1.400000E-01
1.800000E-01  1.800000E-01  1.400000E-01  1.000000E-01  6.000000E-02
2.000000E-02
  
```

```

:=SSTOP
READY.
  
```

ELAPSED TIME IN HUNDREDTHS OF HOURS 003



40

Let the above triangular distribution be approximated by rectangles representing (areas) approximate probabilities of

$$\frac{18}{50}, \frac{14}{50}, \frac{10}{50}, \frac{6}{50}, \frac{2}{50}$$

The analytical probabilities should be (if the above represents both x and y)

<u>Interval</u>	<u>Probability</u>
0 → 2	0.0648
2 → 4	0.1656
4 → 6	0.212
6 → 8	0.2104
8 → 10	0.1672
10 → 12	0.1032
12 → 14	0.0504
14 → 16	0.02
16 → 18	0.0056
18 → 20	0.0008

where for $(0 \leq Z_0 < 10)$

$$\Pr(Z_1 \leq Z_0 \leq Z_2) = \int_0^{Z_0} \int_0^{Z_0-x} \left(\frac{1}{50}\right)^2 (10-x)(10-y) dy dx$$

and for $(10 \leq Z_0 \leq 20)$

$$\Pr(Z_1 \leq Z_0 \leq Z_2) = 1 - \int_{Z_0-10}^{10} \int_{Z_0-x}^{10} \frac{1}{50} (10-x)(10-y) dy dx.$$

$$\Pr (Z_1 \leq Z_0 \leq Z_2) = \int_{Z_0-10}^{10} \int_{Z_0-x}^{10} \frac{1}{50} (10-x) (10-y) dy dx.$$

LOAD LIMITS 06345 15505

:=6,0,2,4,6,8,10.

:=6,0,2,4,6,8,10.

:=11,0,2,4,6,8,10,12,14,16,18,20.

:=.36,.28,.2,.12,.04

:=.36,.28,.2,.12,.04

SUM

1.000000E+00	6.480000E-02	1.656000E-01	2.120000E-01	2.104000E-01
1.672000E-01	1.032000E-01	5.040000E-02	2.000000E-02	5.600000E-03
8.000000E-04				

The probabilities thus tend to "bunch" closer to the lower end of the scale.

For nonlinear frequency distributions, the above procedure is an approximation; however, the finer the grid, the closer the approximation to the true probability.

From these preliminary studies, a program called "BUGS" was written to handle the "mixed" (discrete plus continuous cases) probability addition problem. The program was set up to allow a recursive addition of sample spaces. An example follows which illustrates the addition of four distributions such that the probabilities are discretely defined at 0, 1 and continuously between 1-5 and 5-10. This illustrates the case when the probability of low discrete numbers is important enough to be preserved.

LOAD LIMITS 06622 15505

PROGRAM TO FIND PROBABILITY OF SUMS

NUMBER,POINTS FOR FIRST DENSITY:=6,0,,0,,1,,1,,5,,10.

FIRST SET OF PROBABILITIES:=.6,0,,.2,,.19,,.01

NUMBER,POINTS FOR RESULTING DENSITY:=10,0,,0,,1,,1,,5,,10,,20,,30.
:=40,,50.

NUMBER,POINTS FOR NEXT DENSITY:=6,0,,0,,1,,1,,5,,10.

NEXT SET OF PROBABILITIES:=.7,0,,.21,,.082,,.008

RESULTING PROBABILITIES

4.200000E-01 0.000000E-01 2.660000E-01 2.722675E-01 3.961250E-02
2.120000E-03
CHECK SUM = 1.000000

NUMBER,POINTS FOR NEXT DENSITY:=6,0,,0,,1,,1,,5,,10.

NEXT SET OF PROBABILITIES:=.61,0,,.13,,.17,,.09.

RESULTING PROBABILITIES

2.562000E-01 0.000000E-01 2.168600E-01 3.498813E-01 1.482011E-01
2.857413E-02 2.834743E-04
CHECK SUM = 1.000000

NUMBER,POINTS FOR NEXT DENSITY:=6,0,,0,,1,,1,,5,,10.

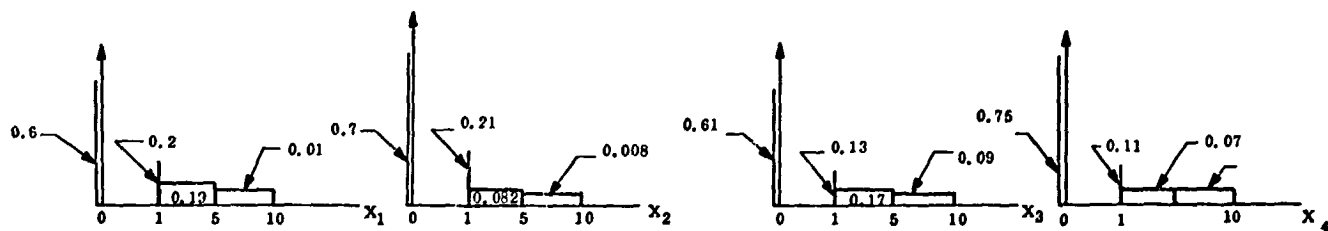
NEXT SET OF PROBABILITIES:=.75,0,,.11,,.07,,.07

RESULTING PROBABILITIES

1.921500E-01 0.000000E-01 1.908270E-01 3.536343E-01 1.984907E-01
6.225241E-02 2.622169E-03 2.343387E-05
CHECK SUM = 1.000000

NUMBER,POINTS FOR NEXT DENSITY:=\$STOP
READY.

INPUT DISTRIBUTIONS

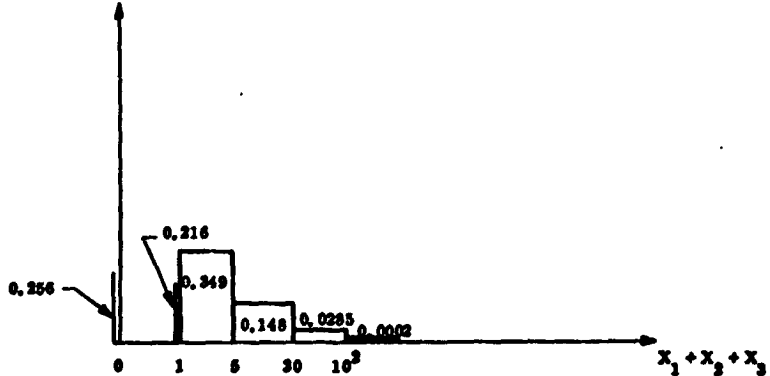


PROBABILITY

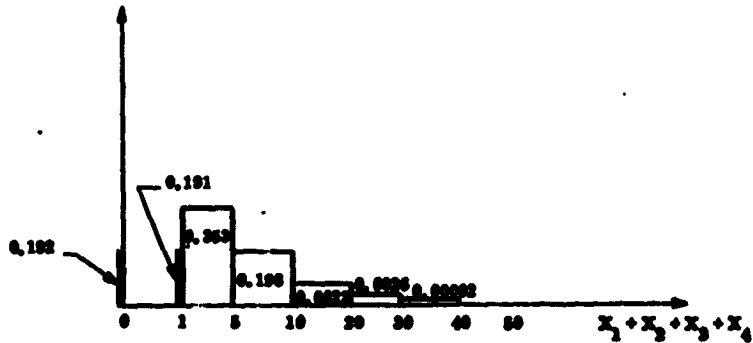


INTERMEDIATE RESULTS

PROBABILITY



PROBABILITY



FINAL RESULTS

44

2.5 DEPENDENT PROBABILITIES

In the calculation of the distribution of a function of other random variables in general, it is assumed that the independent variables are stochastically independent and the joint probability is simply the product.

However, if future needs require the combining of dependent or correlated variables the calculation could be modified.

A simple example can be described.

Suppose we have 2 distributions given by

$$\begin{aligned} p(A) &= \{0.6, 0.4\} \quad \text{at } A = \{0, 1, 2\} \\ \text{and } p(B) &: \{0.1, 0.9\} \quad \text{at } B = \{0, 1, 2\} \end{aligned}$$

Then the distribution of the product of the variables is approximated by

$$p(0 \leq Z < 1) = 0.06 + 0.54 \left(\frac{1}{2}\right) + 0.04 \left(\frac{1}{2}\right) = 0.35$$

$$p(1 \leq Z \leq 2) = 0.54 \left(\frac{1}{2}\right) + 0.04 \left(\frac{1}{2}\right) + 0.36 \left(\frac{1}{3}\right) = 0.41$$

$$p(2 < Z \leq 3) = 0.36 \left(\frac{1}{3}\right) = 0.12$$

$$p(3 \leq Z \leq 4) = 0.36 \left(\frac{1}{3}\right) = 0.12$$

Suppose, however, that the variable B depends on A. How can we describe this? One way is to consider that $p(B)$ is not constant but "varies" with A.

That is, for example, for

$$(0 \leq A \leq 1) : \begin{cases} p(0 \leq B \leq 1) = 0.9 \\ p(1 \leq B \leq 2) = 0.1 \end{cases}$$

but for

$$(1 \leq A \leq 2) : \begin{cases} p(0 \leq B \leq 1) = 0.2 \\ p(1 \leq B \leq 2) = 0.8 \end{cases}$$

This indicates that when A is small, B is likely to be small also and vice versa. In this case the result becomes:

$$p(0 \leq Z \leq 1) = 0.54 + 0.06 \left(\frac{1}{2}\right) + 0.08 \left(\frac{1}{2}\right) = 0.61$$

$$p(1 \leq Z \leq 2) = 0.06 \left(\frac{1}{2}\right) + 0.08 \left(\frac{1}{2}\right) + 0.32 \left(\frac{1}{3}\right) = 0.1766$$

$$p(2 \leq Z \leq 3) = 0.32 \left(\frac{1}{3}\right) = 0.1066$$

$$p(3 \leq Z \leq 4) = 0.32 \left(\frac{1}{3}\right) = 0.1066$$

Thus the independent case can be considered as a special case of dependency where p(B) is unchanged for all intervals of A.

It is anticipated that "DELP" for example, will require this treatment, since pitch down angle and the associated velocity increment conceivably have a correlation.

2.6 MARGINAL PROBABILITY CALCULATION

Probability input appears to be frequently in the form of the probability of surviving a specified event. Also the hypothesis that the probability of an organism surviving remains unchanged from one organism to another seems to be a reasonable assumption in many cases.

The marginal distribution of a random variable x can be written in density function form as

$$f_1(x) = \int f_2(y) f(x|y) dy$$

where y may be thought of as the given random variable and x as the new random variable after suffering the effects of an event.

Numerically this can be approximated by considering

y in interval form

with the following distribution:

$$\Pr \{y_0 < y \leq y_1\} = P_{y1}$$

$$\Pr \{y_1 < y \leq y_2\} = P_{y2}$$

$$\Pr \{y_{n-1} < y \leq y_n\} = P_{yn}$$

Under the binomial hypotheses, the conditional distributions $f(x/y)$ can be generated using the numerical techniques in "BINOM" and "BINOMX."

The resulting marginal distribution which approximates $f_1(x)$ is

$$\Pr \{x_{j-1} < x \leq x_j\} = \sum_{k=1}^n \Pr \{y_{k-1} < y \leq y_k\} \Pr \{x_{j-1} < x \leq x_j \mid \text{given } y_{x-1} < y \leq y_k\}$$

A program called "ICBMAR" was written, which computes $f_1(x)$ over integer values for a selected y values up to or less than 100.

I=55,4,0,1,1.
I=.6,2,0,1,1
I=0,3,7,55

0 7.939340E-01
1 8.766064E-02
2 2.337959E-02
3 1.344830E-02
4 1.607464E-02
5 1.794576E-02
6 1.660133E-02
7 1.291166E-02
8 8.607769E-03
9 4.994631E-03
10 2.552812E-03
11 1.160369E-03
12 4.727429E-04
13 1.737431E-04
14 5.791437E-05
15 1.758881E-05
16 4.885780E-06
17 1.245395E-06
18 2.921297E-07
19 6.320934E-08
20 1.264187E-08
21 2.341087E-09
22 4.020048E-10
23 6.408772E-11
24 9.494477E-12
25 1.308128E-12
26 1.677087E-13
27 2.001462E-14
28 2.223847E-15
29 2.300531E-16
30 2.215326E-17
31 1.985059E-18
32 1.654216E-19
33 1.281043E-20
34 9.210112E-22
35 6.140074E-23
36 3.790169E-24
37 2.162559E-25
38 1.138189E-26
39 5.512596E-28
40 2.450043E-29
41 9.959524E-31
42 3.688712E-32
43 1.239102E-33
44 3.754856E-35
45 1.019837E-36
46 2.463375E-38
47 5.241204E-40
48 9.705970E-42
49 1.540630E-43
50 2.054174E-45
51 2.237662E-47
52 1.912531E-49
53 1.202850E-51
54 4.950000E-54
55 1.000000E-56

1.000000E+00

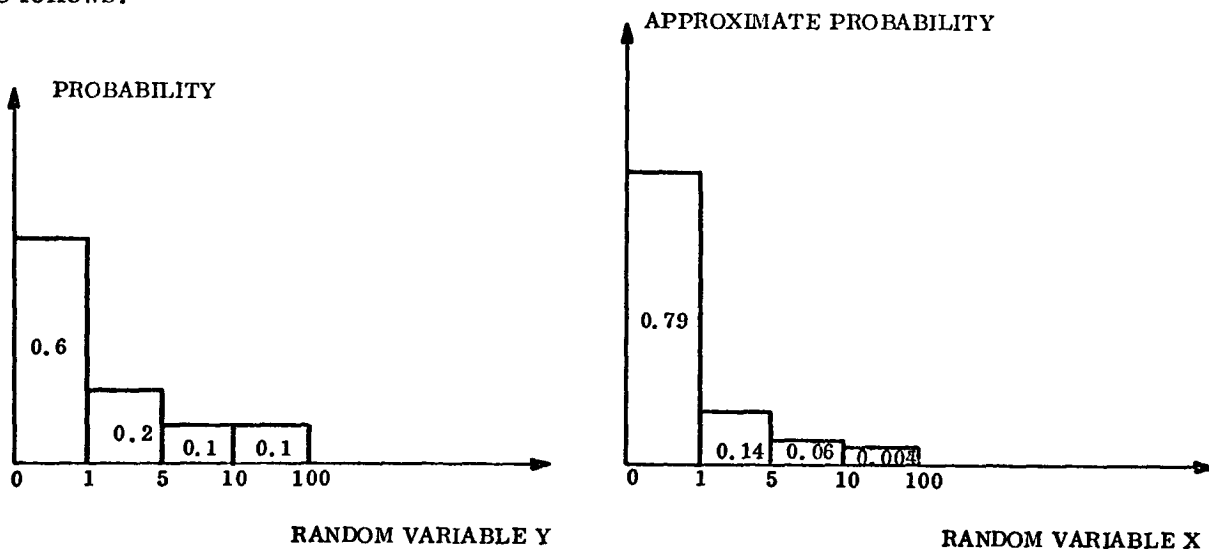
I=SSSTOP
READY.

48

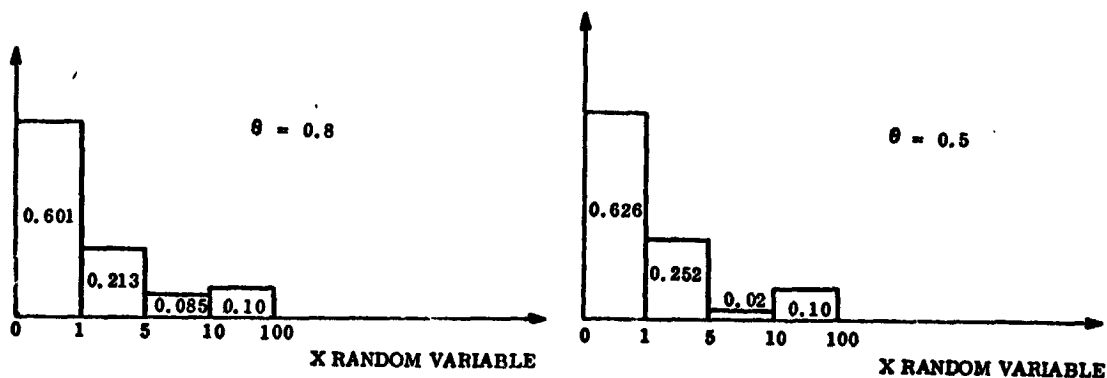
An example was run for $y = \{0, 3, 7, 55\}$ with associated probabilities $\{0.6, 0.2, 0.1, 0.1\}$ and $(0 \leq x \leq 55)$.

The results are shown on the previous page for a probability of survival of 0.1:

If the y values are considered to be mean values for the intervals $(0, 1)$; $(1, 5)$; $(5, 10)$; $(10, 100)$ then the relationship to interval probabilities can be established. If the resulting y probabilities are grouped into intervals then the input and output distributions can be pictured as follows:



As might be expected, the .1 probability has created a "piling up" effect about $X=0$. In the event the probability is large, say .8, the piling effect occurs in two different places and tends to create a bimodal distribution. For the same y distribution two runs were made for a probability of .8 and .5. The approximate results are shown below:



2.7 PERIAPSIS DISTRIBUTION DETERMINATION

Reference 5 contains a set of six curves which relate the function $\Delta p/\Delta V$ to λ where:

Δp = periapsis decrement (km)

ΔV = incremental velocity magnitude ($\frac{m}{s}$)

λ = angle of attack (deg.)

These velocities are small (compared to orbital). This is concerned with ejecta that may enter the Martian atmosphere and thus violate the quarantine. The curves represent a variety of periapsis and apoapsis altitudes. The curves are cosine type and consequently were fitted with a finite Fourier series. The result of the curve fitting appears at the end of this section. The function fitted is of the form:

$$f(\Delta p, \Delta V) = \Delta p / \Delta V \approx \frac{A_0}{Z} \sum_{k=1}^{10} \left\{ A_k \cos\left(\frac{\pi k Z}{9}\right) + B_k \sin\left(\frac{\pi k Z}{9}\right) \right.$$

$$\text{where } Z = \frac{\lambda - 90}{10} \quad (90^\circ \leq \lambda \leq 270^\circ)$$

$f(\Delta p, \Delta V)$ has a double-valued, inverse; however, the point $\lambda = 180$ separates the curve into single valued branches. The interval concept appears to be a satisfactory technique in this case. A two-segment program called "DELP 1" and "DELP 2" was written to compute an approximation to the distribution of $\Delta p = \Delta V \times f(\Delta p, \Delta V)$, given the distributions of λ and ΔV . The program requires the number of the orbit (1 through 6) and proceeds to select the appropriate one. Otherwise, the nature of the input and output is similar to that of the other programs in the package.

A sample run follows. The curve-fitting results are as follows where the cosine terms refer to the A_k and the sine terms refer to the B_k which incidentally are zero due to the cosine nature of the curves.

LOAD LIMITS 07670 15361

PERIAPSIS DISTRIBUTION PROGRAM

ORBIT TYPE(1-6):=1

NUMBER, ANGLE ATTACK VALUES:=7,90.,120.,150.,180.,210.,240.,270.

ANGLE PROBABILITIES, 90. TO 270.:=.29,.2,.01,.01,.2,.29

NUMBER, VALUES FOR PERIAPSIS DECREMENT:=11,0.,1000.,2000.,3000.,
:=4000.,5000.,6000.,7000.,8000.,9000.,10000.

NUMBER, VALUES FOR VELOCITY INCREMENT:=6,0.,200.,400.,600.,800.,1000.

VELOCITY INCREMENT PROBABILITIES:=.2,.2,.2,.2,.2

PERIAPSIS DECREMENT DISTRIBUTION

1.467342E-01 1.037801E-01 1.229073E-01 1.253384E-01 1.261699E-01

1.243205E-01 1.252317E-01 1.249776E-01 4.030254E-05

CHECK SUM = 1.000000

TYPE 1:

DESIRED NUMBER OF HARMONICS TO TRY:=60

DO YOU WISH TO USE MAGNETIC TAPE, TYPE YES OR NO:=NO

TOTAL NUMBER OF DATA POINTS:=18

READ IN DATA POINTS:=.25,1.7,3.,4.2,5.4,6.4,7.1,7.7,8.,8.1,8.
:=7.7,7.1,6.4,5.4,4.2,3.,1.7

HARMONIC	COS TERMS	SINE TERMS	ERROR SS REMOVED
0	1.059444E+01	0.000000E-01	5.050901E+02
1	-3.348694E+00	3.311308E-07	1.009238E+02
2	-7.277126E-01	1.625647E-07	4.766090E+00
3	-2.944445E-01	1.039552E-07	7.802780E-01
4	-1.791110E-01	9.375554E-08	2.887267E-01
5	-1.401124E-01	8.947539E-08	1.766833E-01
6	-1.388889E-01	1.027006E-07	1.736112E-01
7	-9.452752E-02	8.567242E-08	8.041907E-02
8	-7.650987E-02	7.765082E-08	5.268384E-02
9	-9.444452E-02	0.000000E-01	2.006948E-02

TYPE 2:

FINITE FOURIER SERIES

DO YOU WISH TO USE MAGNETIC TAPE, TYPE YES OR NO:=NO

TOTAL NUMBER OF DATA POINTS:=18

READ IN DATA POINTS:=.25,1.6,2.7,3.9,5.,5.9,6.6,7.1,7.4,7.5,7.4
:=7.1,6.6,5.9,5.,3.9,2.7,1.6

HARMONIC	COS TERMS	SINE TERMS	ERROR SS REMOVED
0	9.794444E+00	0.000000E-01	4.316901E+02
1	-3.100466E+00	3.343161E-07	8.651601E+01
2	-6.714439E-01	1.440733E-07	4.057533E+00
3	-2.611111E-01	9.481432E-08	6.136113E-01
4	-1.491711E-01	7.948839E-08	2.002681E-01
5	-1.161407E-01	7.683878E-08	1.213980E-01
6	-1.055556E-01	8.155107E-08	1.002778E-01
7	-1.000602E-01	8.966337E-08	9.010841E-02
8	-9.605177E-02	9.180776E-08	8.303348E-02
9	-9.444452E-02	0.000000E-01	2.006948E-02

TYPE 3:

DESIRED NUMBER OF HARMONICS TO TRY:=GO

DO YOU WISH TO USE MAGNETIC TAPE, TYPE YES OR NO:=NO

TOTAL NUMBER OF DATA POINTS:=18

READ IN DATA POINTS:=.25,1.4,2.5,3.7,4.7,5.6,6.2,6.6,7.,7.1,7.,
:=6.6,6.2,5.6,4.7,3.7,2.5,1.4

HARMONIC	COS TERMS	SINE TERMS	ERROR SS REMOVED
0	9.194444E+00	0.000000E-01	3.804201E+02
1	-2.940966E+00	3.112403E-07	7.784351E+01
2	-6.530806E-01	1.454036E-07	3.838628E+00
3	-2.722222E-01	9.947438E-08	6.669446E-01
4	-1.060668E-01	5.931057E-08	1.012514E-01
5	-1.198379E-01	7.928457E-08	1.292501E-01
6	-7.222224E-02	6.237313E-08	4.694446E-02
7	-5.586342E-02	5.906608E-08	2.808649E-02
8	-9.085276E-02	8.954265E-08	7.420801E-02
9	-7.222228E-02	0.000000E-01	1.173613E-02

TYPE 4:

DESIRED NUMBER OF HARMONICS TO TRY:=GO

DO YOU WISH TO USE MAGNETIC TAPE, TYPE YES OR NO:=NO

TOTAL NUMBER OF DATA POINTS:=18

READ IN DATA POINTS:=.25,2.4,4.4,6.3,8.,9.5,10.7,11.4,11.8,12.,
:=11.8,11.4,10.7,9.5,8.,6.3,4.4,2.4

HARMONIC	COS TERMS	SINE TERMS	ERROR SS REMOVED
0	1.569444E+01	0.000000E-01	1.108420E+03
1	-5.006866E+00	5.580665E-07	2.256183E+02
2	-1.155144E+00	2.570172E-07	1.200922E+01
3	-4.277778E-01	1.519897E-07	1.646945E+00
4	-3.001248E-01	1.479707E-07	8.106742E-01
5	-2.253168E-01	1.341112E-07	4.569089E-01
6	-1.388889E-01	1.043137E-07	1.736112E-01
7	-1.511513E-01	1.250498E-07	2.056203E-01
8	-1.280644E-01	1.155206E-07	1.476045E-01
9	-1.277779E-01	0.000000E-01	3.673617E-02

TYPE 5:

FINITE FOURIER SERIES

DO YOU WISH TO USE MAGNETIC TAPE, TYPE YES OR NO:=NO

TOTAL NUMBER OF DATA POINTS:=18

READ IN DATA POINTS:=.25,2.,4.2,6.1,7.7,9.1,10.3,11.,11.4,11.5,11.4
:=11.,10.3,9.1,7.7,6.1,4.2,2.

HARMONIC	COS TERMS	SINE TERMS	ERROR SS REMOVED
0	1.503889E+01	0.000000E-01	1.017757E+03
1	-4.891183E+00	5.526868E-07	2.153130E+02
2	-1.157197E+00	2.532923E-07	1.205195E+01
3	-4.500000E-01	1.591590E-07	1.822500E+00
4	-3.137546E-01	1.549004E-07	8.859775E-01
5	-1.722799E-01	1.057807E-07	2.671232E-01
6	-9.444444E-02	7.599485E-08	8.027777E-02
7	-8.653781E-02	7.955296E-08	6.739914E-02
8	-7.904804E-02	7.977436E-08	5.623733E-02
9	-5.000005E-02	0.000000E-01	5.625011E-03

TYPE 6:

DESIRED NUMBER OF HARMONICS TO TRY:=60

DO YOU WISH TO USE MAGNETIC TAPE, TYPE YES OR NO:=NO

TOTAL NUMBER OF DATA POINTS:=18

READ IN DATA POINTS:=.25,1.3,3.9,5.7,7.6,8.8,9.9,10.7,11.1,11.3
:=11.1,10.7,9.9,8.8,7.6,5.7,3.9,1.3

HARMONIC	COS TERMS	SINE TERMS	ERROR SS REMOVED
0	1.439444E+01	0.000000E-01	9.324001E+02
1	-4.944771E+00	5.681179E-07	2.200569E+02
2	-1.200388E+00	2.695222E-07	1.296838E+01
3	-4.944444E-01	1.717053E-07	2.200278E+00
4	-2.284768E-01	1.161753E-07	4.698150E-01
5	-1.005758E-01	6.827849E-08	9.103944E-02
6	-7.222217E-02	6.183543E-08	4.694438E-02
7	-3.798681E-02	4.629503E-08	1.298698E-02
8	7.886461E-02	-3.737432E-08	5.597663E-02
9	1.055556E-01	0.000000E-01	2.506946E-02

2.8 MARS ORBIT TIME AND $M/C_d A$ DISTRIBUTION

Reference 5 contains two curves relating the quantity $T/h_a (M/C_d A)$ to periapsis altitude (all distances in km). The terms are: T , orbital lifetime (years); h_a , apoapsis altitude for the six types of orbits in the periapsis section; and $M/C_d A$, drag parameter.

The independent variable $P = P_a - \Delta P$ where P_a is the periapsis altitude for the type of orbit under consideration and ΔP is the random variable whose distribution has been determined in the periapsis section. Each curve represents extremes in the VM-3 atmosphere variation. The curves were fit by fitting orthogonal polynomials to $\ln (T/h_a (M/C_d A))$ VS. P . That is, each curve was approximated by

$$f(T, h_a, M/C_d A) = T/h_a (M/C_d A) = e^{\left(\sum^4 B_j \Phi_j (P)\right)}$$

where $\Phi_j (P)$ are orthogonal polynomials of degree j . The shape of the curves indicates that transforming to \sqrt{P} would help the approximation, and so this will be attempted at a later date. The present results do look satisfactory, however. The curve-fitting results are shown at the end of this section.

Since the curves are monotonic, the interval technique should be effective. The programs titled "TIMF1" and "M/C_d A1" were written to provide the distributions of T and $M/C_d A$, respectively. Obviously, the former approximates the density of $T = h_a \times (M/C_d A) \times f(T, h_a, M/C_d A)$, while the latter approximates $T = h_a \times \frac{1}{f(T, h_a, M/C_d A)}$.

In "TIME" the orbit type (1 through 6) and atmosphere type (1 or 2, where 1 is the upper curve and 2 is the lower curve) are entered initially. The associated h_a and P_a are printed out as a check. Following this the grid and probabilities for P and $(M/C_d A)$ are entered, including the output (T) grid in the usual way. Note that T , $M/C_d A$, and P are all considered as random variables in each case. A typical run follows.

READY.

SRUN
WAIT.

LOAD LIMITS 07440 15311

TIME IN MARS ORBIT PROGRAM

ORBIT TYPE(1-6), ATMOS TYPE(1 OR 2):=1,1

APOAPSIS(KM) PERIAPSIS(KM)
1.000000E+04 1.000000E+03
NUMBER, PERIAPSIS VALUES:=3,200.,600.,1000.

PERIAPSIS PROBABILITIES:=.7,.3

NUMBER, VALUES FOR TIME IN ORBIT:=6,0.,5.,10.,100.,1000.,2000.

NUMBER, VALUES FOR DRAG PARAMETER:=3,1.E-5,1.E-4,1.E-3

DRAG PARAMETER PROBABILITIES:=.4,.6

TIME IN ORBIT DISTRIBUTION
2.218431E-01 1.235655E-01 4.544181E-01 1.515289E-01 4.864454E-02
CHECK SUM = 1.000000

READY.

SRUN
WAIT.

LOAD LIMITS 07440 15311

M/CDA DISTRIBUTION PROGRAM

ORBIT TYPE(1-6), ATMOS TYPE(1 OR 2):=1,1

APOAPSIS(KM) PERIAPSIS(KM)
1.000000E+04 1.000000E+03
NUMBER, PERIAPSIS VALUES:=3,200.,600.,1000.

PERIAPSIS PROBABILITIES:=.7,.3

NUMBER, VALUES FOR M/CDA:=20,0.,10.,100.,1000.,1.E4,1.E5,1.E6
:=1.E7,1.E8,1.E9,1.E10,1.E20,1.E30,1.E40,1.E50,1.E60,1.E70
:=1.E71,1.E72,1.E72

NUMBER, VALUES FOR TIME IN ORBIT:=6,0.,1.,2.,3.,5.,10.

TIME IN ORBIT PROBABILITIES:=.2,.2,.2,.2,.2

M/CDA DISTRIBUTION
2.336721E-01 2.852110E-01 3.811170E-01
CHECK SUM = 1.000000

The curve-fitting results for the pair of curves is shown below:

ORTHOGONAL POLYNOMIAL CURVE FITTING

DO YOU WISH TO USE MAGNETIC TAPE, TYPE YES OR NO:=NO

TYPE NUMBER OF POINTS, MAXIMUM DEGREE:=9,8

TYPE IN DEPENDENT DATA:=-13.81551,-5.298317,-1.609438,.6931472
:2.079442,2.995732,3.912023,4.382027,5.010635

TYPE IN INDEPENDENT DATA:=200.,300.,400.,500.,600.,700.,800.,900.,1000.

TYPE IN WEIGHTS:=1.,1.,1.,1.,1.,1.,1.,1.,1.

DEPENDENT DATA MEAN

-1.833621E-01

DEGREE	ALPHA	BETA	COEFF	SSR
1	.3000000E+02	0.000000E-01	1.961519E-02	2.308533E+02
2	6.000000E+02	6.666667E+04	-4.065703E-05	5.091223E+01
3	6.000000E+02	5.133333E+04	8.695401E-08	1.077896E+01
4	6.000000E+02	4.628571E+04	-1.711093E-10	1.722574E+00
5	6.000000E+02	4.126984E+04	3.727533E-13	2.890057E-01
6	6.000000E+02	3.535354E+04	-6.375616E-16	2.394565E-02
7	6.000000E+02	2.832168E+04	5.762255E-18	3.932047E-02
8	6.000000E+02	2.010256E+04	4.914316E-21	3.050624E-04

DESIRED NUMBER OF POLYNOMIALS TO TRY:=4

WHICH ONES:=1,2,3,4

INPUT:=200.

PREDICTED VALUE -1.369558E+01
INPUT:=300.

PREDICTED VALUE -5.670175E+00
INPUT:=400.

PREDICTED VALUE -1.343066E+00
INPUT:=500.

PREDICTED VALUE 8.341239E-01
INPUT:=600.

PREDICTED VALUE 1.999112E+00
INPUT:=700.

PREDICTED VALUE 2.878955E+00
INPUT:=800.

PREDICTED VALUE 3.790044E+00
INPUT:=900.

PREDICTED VALUE 4.638110E+00
INPUT:=1000.

PREDICTED VALUE 4.918221E+00
INPUT:=1.E75

DESIRED NUMBER OF POLYNOMIALS TO TRY:=30

DO YOU WISH TO USE MAGNETIC TAPE, TYPE YES OR NO:=NO

TYPE NUMBER OF POINTS, MAXIMUM DEGREE:=9,8

TYPE IN DEPENDENT DATA:=-13.81551,-8.111728,-5.298317,-3.912023
:-2.65926,-1.609438,-.5108256,0.,.6931472

TYPE IN INDEPENDENT DATA:=200.,300.,400.,500.,600.,700.,800.,900.,1000.

TYPE IN WEIGHTS:=1.,1.,1.,1.,1.,1.,1.,1.,1.

DEPENDENT DATA MEAN
-3.913773E+00

DEGREE	ALPHA	BETA	COEFF	SSR
1	6.000000E+02	0.000000E-01	1.570790E-02	1.487428E+02
2	6.000000E+02	6.666667E+04	-2.496592E-05	1.919755E+01
3	6.000000E+02	5.133333E+04	5.334886E-08	4.057402E+00
4	6.000000E+02	4.628571E+04	-1.370223E-10	1.104620E+00
5	6.000000E+02	4.126984E+04	2.781683E-13	1.609454E-01
6	6.000000E+02	3.535354E+04	4.881966E-16	1.404015E-02
7	6.000000E+02	2.832168E+04	6.220064E-19	4.581667E-04
8	6.000000E+02	2.010256E+04	3.018112E-20	1.150623E-02

DESIRED NUMBER OF POLYNOMIALS TO TRY:=4

WHICH ONES:=1,2,3,4

INPUT:=200.

PREDICTED VALUE -1.375220E+01
INPUT:=300.

PREDICTED VALUE -8.267269E+00
INPUT:=400.

PREDICTED VALUE -5.298967E+00
INPUT:=500.

PREDICTED VALUE -3.705065E+00
INPUT:=600.

PREDICTED VALUE -2.672190E+00
INPUT:=700.

PREDICTED VALUE -1.715821E+00
INPUT:=800.

PREDICTED VALUE -6.802930E-01
INPUT:=900.

PREDICTED VALUE 2.612080E-01
INPUT:=1000.

PREDICTED VALUE 6.066411E-01
INPUT:=\$STOP

2.9 HELIOCENTRIC TRANSFER CASE

Reference 5 contains information on the various effects on the Mars impact miss distance during the transfer orbit phase. This report contains four curves which relate the four following quantities to time in days to intercept.

- a. In-plane miss distance due to tangential component of ejection velocity.

$$T_1 \Delta V_T = f_1(t) \quad (\text{km} / \frac{\text{m}}{\text{s}})$$

- b. In-plane miss distance due to normal ejection velocity component.

$$T_2 / \Delta V_N = f_2(t) \quad (\text{km} / \frac{\text{m}}{\text{s}})$$

The results here were multiplied by 10^3 to obtain the necessary units.

- c. Radiation pressure perturbation to transfer trajectories.

$$T_3 \left(\frac{M}{C_d A} \right) = f_3(t) \quad \text{km} / \left(\frac{\text{slugs}}{\text{ft}^2} \right)$$

Two curves (Type I, 1973 and Type II, 1975)

- d. Out-of-plane component of particle miss distance caused by out-of-plane component of ejection velocity.

$$R / \Delta V_R = f_4(t) \quad (\text{km} / \frac{\text{m}}{\text{s}})$$

Two curves (Type I, 1973 and Type II, 1975). The results here were multiplied by 10^3 to obtain the necessary units.

Thus the four random variables (ΔV_T , ΔV_N , $M/C_d A$, and DV_R) contribute to the Mars quarantine area miss distance, including a bias deliberately programmed into the guidance system.

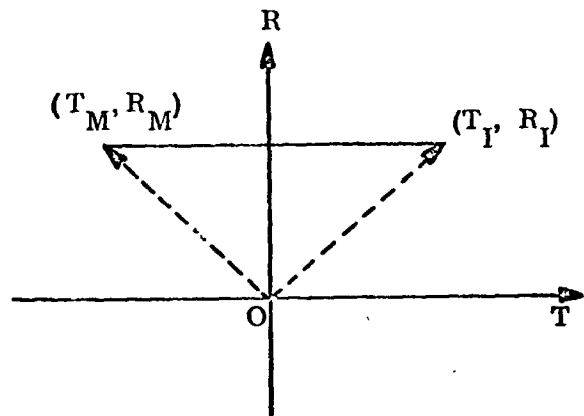
Denoting R and T as the out-of-plane and in-plane components in the impact plane, the impact point (T_I, R_I) components are given

$$\text{by } T_I = T_1 + T_2 + T_3$$

$$R_I = R.$$

The miss distance from Mars (T_m, R_m) is given by

$$d = \sqrt{(T_I - T_m)^2 + (R_I - R_m)^2} \text{ (kilometers)}$$



It is the job of the two segments "HELIO 1" and HELIO 2" to approximate the distribution

of d , given the above-described random

variables. The four (actually six) curves were fit in the following ways:

- a. Orthogonal polynomials were used to fit $\ln f_1(t)$ VS \sqrt{t} .
- b. Orthogonal polynomials were used to fit $10^3 \times f_2(t)$ VS. t
- c. Orthogonal polynomials were used to fit $\ln f_3(t)$ VS. \sqrt{t} (for both curves).
- d. Orthogonal polynomials were used to fit $10^3 \times f_4(t)$ VS t (for both curves).

The results of the fits follow for all six curves. The procedure is fairly simple. The user inputs the four grids and associated probabilities and the output (d) grid. He also must provide: days to impact, orbit type (1 or 2-needed for $f_3(t)$, $f_4(t)$, and the Mars bias coordinates in the impact plane (in km.) The program then samples the appropriate curves and calculates intervals for:

- a. $T_1 = \Delta V_T \times f_1(t)$
- b. $T_2 = \Delta V_N \times 10^3 \times f_2(t)$
- c. $T_3 = f_3(t) / M/C_d A$
- d. $R = \Delta V_R \times 10^3 \times f_4(t)$

Thus for n_1, n_2, n_3, n_4 intervals, the program forms $n_1 \times n_2 \times n_3 \times n_4$ intervals and computes $d \sqrt{(T_1 + T_2 + T_3 - T_m)^2 + (R - R_m)^2}$ for each interval. The probability associated with each interval is the product of the probability for each variable for the particular intervals concerned.

ORTHOCONAL POLYNOMIAL CURVE FITTING

DO YOU WISH TO USE MAGNETIC TAPE, TYPE YES OR NO:=NO $f_n(\frac{T_1}{\Delta V_T}) \sim \frac{km}{m/s}$

TYPE NUMBER OF POINTS, MAXIMUM DEGREE:=13, 12

TYPE IN DEPENDENT DATA:=10.81978, 10.73204, 10.63586, 10.18867
 :=9.21034, 8.922658, 6.907755, 5.857933, 4.434382, 3.931826
 :=2.944439, 2.302585, 0.

TYPE IN INDEPENDENT DATA:=18.70829, 17.32051, 15.81139
 :=14.14214, 12.24745, 10., 7.071068, 5., 3.535534, 2.5, 1.767767
 :=1.251998, 0.

TYPE IN WEIGHTS:=1., 1., 1., 1., 1., 1., 1., 1., 1., 1., 1., 1., 1., 10.

DEPENDENT DATA MEAN
 3.949467E+00

DEGREE	ALPHA	BETA	COEFF	SSR
1	4.970734E+00	0.000000E-01	6.436589E-01	3.748817E+02
2	1.128421E+01	4.113012E+01	-3.821528E-02	2.392741E+01
3	9.779519E+00	1.810669E+01	1.998154E-03	1.639684E+00
4	9.675320E+00	2.506573E+01	-2.700023E-04	6.134089E-01
5	9.866702E+00	2.048864E+01	2.481163E-05	1.108942E-01
6	9.626673E+00	2.140835E+01	-5.069287E-06	8.658075E-02
7	9.838191E+00	1.870377E+01	4.697225E-07	1.422646E-02
8	9.183295E+00	1.913756E+01	2.604144E-07	7.278603E-02
9	9.748145E+00	1.664577E+01	2.243741E-08	8.901758E-03
10	8.043692E+00	1.640132E+01	-1.189293E-08	3.132336E-02
11	1.002555E+01	1.378526E+01	-6.717493E-09	1.244618E-01
12	4.868961E+00	1.136606E+01	-5.608993E-09	6.941599E-02

ORTHOGONAL POLYNOMIAL CURVE FITTING

DO YOU WISH TO USE MAGNETIC TAPE, TYPE YES OR NO:=NO $10^3 \times \frac{T_2}{\Delta V_N} \sim \frac{km}{m/s}$

TYPE NUMBER OF POINTS, MAXIMUM DEGREE:=8, 7

TYPE IN DEPENDENT DATA:=-4.6, -.8, 3., 6.6, 8.2, 7.3, 4.2, 0.

TYPE IN INDEPENDENT DATA:=350., 300., 250., 200., 150., 100., 50., 0.

TYPE IN WEIGHTS:=1., 1., 1., 1., 1., 1., 1., 1.

DEPENDENT DATA MEAN
 2.987500E+00

DEGREE	ALPHA	BETA	COEFF	SSR
1	1.750000E+02	0.000000E-01	-1.707143E-02	3.060054E+01
2	1.750000E+02	1.312500E+04	-3.183333E-04	1.064029E+00
3	1.750000E+02	1.000000E+04	5.595960E-07	2.906400E+00
4	1.750000E+02	8.839286E+03	3.924243E-09	1.088977E+00
5	1.750000E+02	7.619048E+03	-1.466667E-11	9.363094E-02
6	1.750000E+02	6.155303E+03	-1.288889E-13	3.185605E-02
7	1.750000E+02	4.405594E+03	-9.904750E-16	4.431807E-03

ORTHOGONAL POLYNOMIAL CURVE FITTING

DO YOU WISH TO USE MAGNETIC TAPE, TYPE YES OR NO:=NO $\ln(T_3 \frac{M}{C_d A})$

TYPE NUMBER OF POINTS, MAXIMUM DEGREE:=7, 6

Type I (1973)

TYPE IN DEPENDENT DATA:=8.373322, 6.907755, 6.39693
:=4.447346, 2.928524, 2.079442, 0.

TYPE IN INDEPENDENT DATA:=14.14214, 12.24745, 10., 7.071062
:=5., 3.162278, 0.

TYPE IN WEIGHTS:=1., 1., 1., 1., 1., 1., 100.

DEPENDENT DATA MEAN

2.237106E-01

DEGREE	ALPHA	BETA	COEFF	SSD
1	4.270088E-01	0.000000E-01	5.972080E-01	1.818451E+02
2	1.111457E+01	4.809924E+00	-5.188701E-03	1.184448E-01
3	8.302233E+00	8.628762E+01	-2.828954E-04	3.424910E-03
4	8.578036E+00	9.727467E+00	3.301192E-04	4.112717E-02
5	7.382239E+00	8.833383E+00	3.147279E-04	2.759230E-01
6	8.428462E+00	7.368716E+00	9.506986E-06	1.278175E-02

ORTHOGONAL POLYNOMIAL CURVE FITTING

DO YOU WISH TO USE MAGNETIC TAPE, TYPE YES OR NO:=NO $\ln(T_3 \frac{M}{C_d A})$

TYPE NUMBER OF POINTS, MAXIMUM DEGREE:=13, 9

Type II (1975)

TYPE IN DEPENDENT DATA:=8.207947, 7.839334, 7.192234, 6.220476
:=6.51247, 5.911141, 4.312141, 2.995722, 2.012895
:=0.

TYPE IN INDEPENDENT DATA:=13.70829, 17.32251, 15.31132, 14.14214
:=12.24745, 10., 7.071062, 5., 3.162278, 0.

TYPE IN WEIGHTS:=1., 1., 1., 1., 1., 1., 1., 1., 1., 100.

DEPENDENT DATA MEAN

4.751153E-01

DEGREE	ALPHA	BETA	COEFF	SSD
1	9.492030E-01	0.000000E-01	4.797478E-01	3.076736E+02
2	1.439388E+01	1.226415E+01	-1.435841E-02	3.523358E+00
3	1.094751E+01	1.282067E+01	1.445613E-04	7.015002E-03
4	1.263706E+01	1.959507E+01	1.624449E-04	1.484310E-01
5	1.075831E+01	1.674908E+01	4.462266E-06	1.053010E-03
6	1.032720E+01	1.746049E+01	-6.352601E-06	7.125051E-02
7	1.095632E+01	1.544923E+01	-5.471351E-07	5.610214E-03
8	1.046540E+01	1.269522E+01	-1.775542E-07	0.023370E-02
9	1.089951E+01	1.322024E+01	-7.256362E-09	6.021747E-05

ORTHOGONAL POLYNOMIAL CURVE FITTING

DO YOU WISH TO USE MAGNETIC TAPE, TYPE YES OR NO:=NO

TYPE NUMBER OF POINTS, MAXIMUM DEGREE:=5,4

$$10^3 \times \frac{R}{\Delta}$$

TYPE IN DEPENDENT DATA:=3.3,7.4,6.8,3.4,0.

Type I (1978)

TYPE IN INDEPENDENT DATA:=200.,150.,100.,50.,0.

TYPE IN WEIGHTS:=1.,1.,1.,1.,50.

DEPENDENT DATA MEAN

3.870370E-01

DEGREE	ALPHA	BETA	COEFF	SSR
1	9.259259E+00	0.000000E-01	3.448158E-02	8.366891E+01
2	1.585039E+02	1.303155E+03	-4.781210E-04	2.921414E+01
3	1.227774E+02	1.816049E+03	-3.141853E-06	2.466142E+00
4	1.080396E+02	1.954920E+03	6.000009E-09	1.173576E-02

DO YOU WISH TO USE MAGNETIC TAPE, TYPE YES OR NO:=NO

TYPE NUMBER OF POINTS, MAXIMUM DEGREE:=8,7

$$10^3 \times \frac{R}{\Delta}$$

Type II (1975)

TYPE IN DEPENDENT DATA:=-3.,3.3,6.9,7.7,6.6,4.9,2.7,0.

TYPE IN INDEPENDENT DATA:=350.,300.,250.,200.,150.,100.,50.,0.

TYPE IN WEIGHTS:=1.,1.,1.,1.,1.,1.,1.,1.

DEPENDENT DATA MEAN

3.637500E+00

DEGREE	ALPHA	BETA	COEFF	SSR
1	1.750000E+02	0.000000E-01	-2.595238E-03	7.070024E-01
2	1.750000E+02	1.312500E+04	-2.902381E-04	8.845006E+01
3	1.750000E+02	1.000000E+04	-8.343434E-07	6.469947E+00
4	1.750000E+02	8.839286E+03	-8.636362E-10	5.274349E-02
5	1.750000E+02	7.619048E+03	1.610256E-11	1.128617E-01
6	1.750000E+02	6.155303E+03	7.555557E-14	1.094697E-02
7	1.750000E+02	4.405594E+03	-9.396823E-16	3.988926E-03

GK

\$LOAD HELIO1,HELIO2

LOAD LIMITS 11643 13611

HELIOCENTRIC ORBIT PROBABILITY PROGRAM

DAYS TO IMPACT, ORBIT TYPE,T,R MARS :=3,1,0.,0.

NUMBER, TANGENTIAL VELOCITY VALUES:=3,0.,5.,10.

TANGENTIAL PROBABILITIES:=.8,.2

NUMBER, NORMAL VELOCITY VALUES:=3,0.,5.,10.

NORMAL PROBABILITIES:=.7,.3

NUMBER, M/CDA VALUES:=4,1.E-1,1.E-2,1.E-3,1.E-4

M/CDA PROBABILITIES:=.7,.2,.1

NUMBER, O-O-P VELOCITY VALUES:=3,0.,5.,10.

O-O-L PROBABILITIES:=.6,.4

NUMBER, MISS DISTANCE VALUES:=10,0.,100.,1000.,1.E4,1.E5,1.E6
:=1.E7,1.E8,1.E9,1.E10

T1/DVT	T2/DVN	T3(M/CDA)	R/DVR	C-F
1.593031E+01	2.415908E+02	2.528376E+00	1.690686E+02	
IN-PLANE MISS DISTANCE PROBABILITIES				
1.014702E-02	1.921751E-01	7.269995E-01	7.067839E-02	
CHECK SUM = 1.000000				

DAYS TO IMPACT, ORBIT TYPE,T,R MARS :=\$STOP
READY.

ADDENDUM TO HELIO1, HELIO2

The segments HELIO1, HELIO2 perform as described in PIR 5540-41.

A new program, to be loaded as HELIO3, HELIO2, was written to allow the user to input ejection velocity magnitude and two angles along with drag parameter. These four quantities are considered to be stochastically independent.

Define a local axis system as N, T, R where N is the local normal of the velocity vector, T is the local tangent of the velocity vector, and R is the out of (transfer) plane component.

It is along these three axes that the "old" program HELIO1, HELIO2 considered as its basic input.

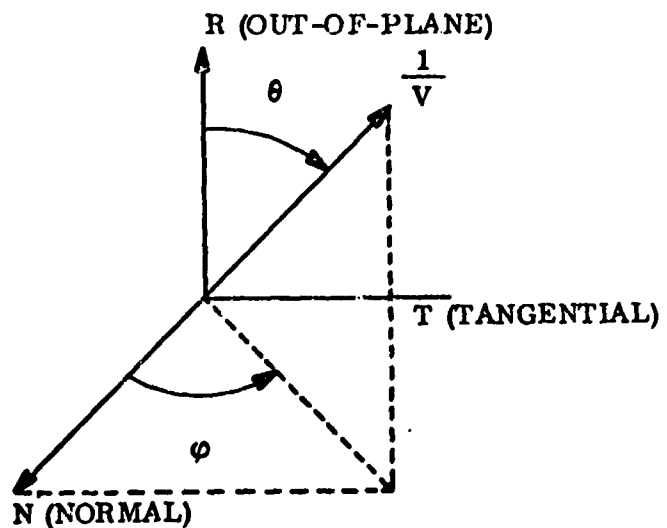
Define

V, magnitude of velocity ($\frac{m}{s}$)

θ , polar angle (deg)

φ , N-T plane angle (deg)

β , drag term ($\frac{\text{slugs}}{\text{ft}^2}$)
($M/C_d A$)



The miss distance from the center of Mars is calculated as

$$d = \sqrt{(T-T_M)^2 + (R-R_M)^2} \text{ (km)}$$

where $T = T_1 + T_2 + T_3$

$$= C_1 V \sin \theta \sin \varphi + C_2 V \sin \theta \cos \varphi + C_3 / \beta$$

$$R = C_4 V \cos \theta$$

and

$$C_1 = T_1/\Delta V_T$$

$$C_2 = T_2/\Delta V_N$$

$$C_3 = T_3\beta$$

$$C_4 = R/\Delta V_R$$

which are curve-fitted results of curves supplied by D. A. Korenstein given as a function of "TIME" (days to intercept).

T_M and R_M are the coordinates of the center of Mars in the impact plane. (T, transfer plane direction and R, out of transfer plane direction.)

Note that the T_1 , T_2 , T_3 components of velocity are not independent and involve a complete different numerical process as performed in HELIO1, HELIO2.

The numerical technique involves calculating all 16 possible d's for each combination of random variable values (V , θ , φ , β), choosing the minimum and maximum, and loading the associated probabilities by the technique described in the writeup of "PLOAD."

Thus, for n_1 values of V , n_2 values of θ , n_3 values of φ , and n_4 values of β the program must calculate a total of $2^4 \times (n_1-1) \times (n_2-1) \times (n_3-1) \times (n_4-1)$ times. Also $(n_1-1) \times (n_2-1) \times (n_3-1) \times (n_4-1)$ intervals are loaded onto the "d" grid as in the usual manner.

SAMPLE PROBLEM

A time of 5 days to impact was chosen for a TYPE1 orbit and Mars coordinates of (-10., -10).

SSTOP
READY.

CHECK CASE
12/15/66

SLOAD HELI03, HELI02

LOAD LIMITS 11521 13611

HELIOCENTRIC ORBIT PROBABILITY PROGRAM

DAYS TO IMPACT, ORBIT TYPE, T, R MARS :=5, 1, -10., -10.

NUMBER, VELOCITY MAG VALUES(M/S):=3, 0., 10., 20.

VELOCITY MAG PROBABILITIES:=.8, .2

NUMBER, POLAR ANGLE VALUES(DEG):=2, 5., 15.

POLAR ANGLE PROBABILITIES:=1.

NUMBER, N-T PLANE ANGLE VALUES(DEG):=2, 5., 15.

N-T PLANE ANGLE PROBABILITIES:=1.

NUMBER, M/CDA VALUES(SLUGS/FT*FT):=2, 1.E-3, 1.E-2

M/CDA PROBABILITIES:=1.

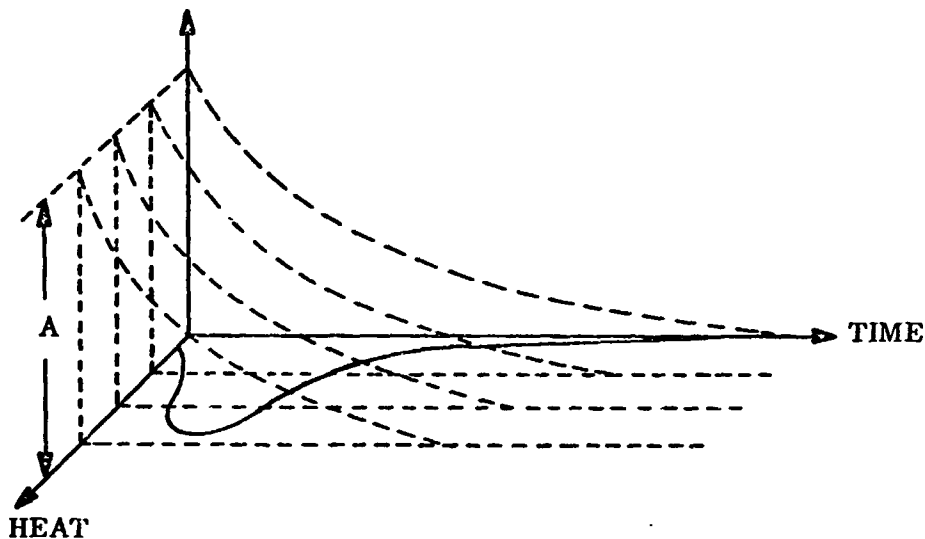
NUMBER, MISS DISTANCE VALUES (KM):=10, 0., 1000., 2000., 3000., 4000.,
:=5000., 10000., 1.E5, 1.E6, 1.E7

T1/DVT	T2/DVN	T3(M/CDA)	R/DVR	C-F
2.991216E+01	4.263029E+02	3.496503E+00	2.860507E+02	
***** MISS DISTANCE PROBABILITIES *****				
1.019843E-01	1.592980E-01	1.612878E-01	1.992267E-01	1.992267E-01
1.789764E-01				
CHECK SUM = 1.000000				

2.10 ENTRY SURVIVAL PROBABILITY

Reference (1) contains a description of the parameterization of the estimated effect of heat-time on viable organisms entering the Martian atmosphere.

NO. OF SURVIVING ORGANISMS



The above diagram illustrates the process. If we have "A" organisms to start with, the die off will proceed (negative exponential) as is shown in the dotted curves at constant temperatures. The history of an entering particle may suffer a heat-time curve in the heat-time plane as illustrated.

Reference 1 develops a justification for computing a particular index of the particle history called the lethality integral (I_L). Once I_L is computed, the probability of an organism surviving (to some indicated percentage) is

$$\left(\frac{1}{A} \right)^{I_L}$$

Thus if I_L is a random variable itself, the probability of survival can be estimated by

$$\text{pr (survival)} = \sum_j \text{pr} (I_L = I_{L_j}) \left(\frac{1}{A} \right)^{I_{L_j}}$$

In Reference 1, I_L is considered to be a function of several parameters. In particular, four seem to be the most important:

- ϵ = emissivity
- ν = initial entry velocity
- γ = initial entry angle
- Z = drag parameter

M. A. Martin has demonstrated that the relationship:

$$\rho n_{1\epsilon} I_L = 3.34036 - 5.34036 \left(\frac{\xi}{\xi_2} \right)$$

where

$$\xi_2 = k_1 + k_2 \bar{Z} + k_3 \bar{\gamma} + k_4 \bar{\nu} + k_5 \bar{Z} \bar{\gamma} + k_6 \bar{\gamma} \bar{\nu} + k_7 \bar{\nu} \bar{Z} + k_8 \bar{Z} \bar{\gamma} \bar{\nu}$$

where

$$\bar{Z} = Z \times 10^4$$

$$\bar{\gamma} = \left(\frac{90 - \gamma_2}{100} \right)$$

$$\bar{\nu} = \left(\frac{\nu}{10^4} \right)^3$$

is a satisfactory form in his preliminary studies from available data. Appropriately, the program "LID" was written to compute the probability distribution of I_L . The input is by the same method of providing grid intervals and probabilities used in other programs.

NUMBER, BALLISTIC COEFFICIENTS
:=3,4.E-5,22.E-5,4.E-4

BALLISTIC PROBABILITIES
:=.8,.2

NUMBER, INITIAL ENTRY ANGLES
:=3,5.,10.,25..

ENTRY ANGLES PROBABILITIES
:=.5,.5

NUMBER, INITIAL PARTICLE VELOCITY
:=3,12000.,13000.,14000..

PARTICLE VELOCITY PROBABILITIES
:=.5,.5

NUMBER, EMISSIVITIES
:=3.,2.,.3.,.4

EMISSIVITY PROBABILITIES
:=.7,.3

NUMBER, LETHALITY INTEGRAL
:=10,1.E-4,1.E-3,1.E-2,1.E-1,1.,2.,5.,10.,1.E5--- 100.,1.E5

***LETHALITY PROBABILITIES

2.934343E-04	2.648213E-03	2.649075E-02	1.338197E-01	4.328347E-02
1.298504E-01	1.236249E-01	3.835066E-01	1.564824E-01	

07704 EOT
SUM = 1.000000E+00

NUMBER, BALLISTIC COEFFICIENTS
:=SSTOP
READY..

2.11 M/CDA SURVIVAL PROBABILITIES

According to Reference 7, one method of determining the distribution of M/CDA that enters the atmosphere is to generate a distribution of upper limits on M/CDA entering the atmosphere (M/CDA program).

This distribution is then merged with the given M/CDA distribution to determine the surviving distribution of M/CDA that enters the atmosphere.

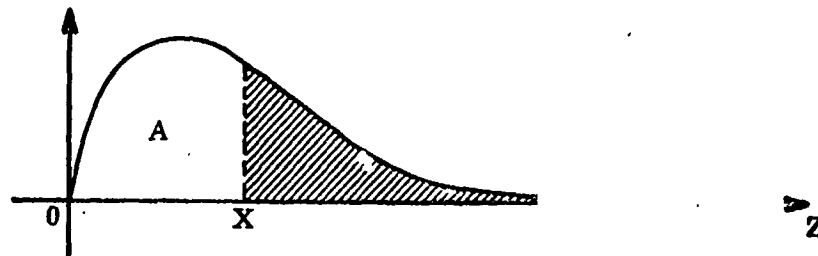
Define

Z = original a priori random variable

Z_{ul} = upper limit random variable

Z_a = resulting "a posteriori" random variable

When a value for $Z_{ul} = \alpha$ is given, the conditional distribution for Z



can be found by dividing the modified area A. That is the density of Z is modified to form

$$f(Z | Z_{ul} = \alpha) = \frac{f(Z)}{A}.$$

The final distribution becomes:

$$f(Z_A) = \sum_{Z_{ul}} g(Z_{ul}) f(Z | Z_{ul})$$

The numerical procedure consists of reading the density for Z and Z_{ul} in interval probability form and generating the distributing of Z_A in the same form by calculating the summation described above. A program now exists on the DSCS to perform this calculation.

The program is known as "LIMIT".

To use the program, provide first the upper limit points and related probabilities; next the M/C_D A points and probabilities and finally the desired put points for the resulting marginal distribution.

```

PROGRAM TO COMBINE UPPER LIMIT AND M/CDA VARIABLES.
NUMBER,UPPER LIMIT POINTS:=5,0,1,2,3,4.
UPPER LIMIT PROBABILITIES:=.25,.25,.25,.25
NUMBER,M/CDA POINTS:=5,0,1,2,3,4.
M/CDA PROBABILITIES:=.6,2,1,1
NUMBER,OUTPUT POINTS:=10,0,1,2,3,4,5,10,20,30,50.

FIRST      LAST      PROBABILITY
0.00000E+01  1.00000E+00  7.541667E-01
1.00000E+00  2.00000E+00  1.489556E-01
2.00000E+00  3.00000E+00  5.27777E-02
3.00000E+00  4.00000E+00  2.50000E-02
4.00000E+00  5.00000E+00  8.00000E-03
5.00000E+00  1.00000E+01  0.00000E+00
1.00000E+01  2.00000E+01  0.00000E+00
2.00000E+01  3.00000E+01  0.00000E+00
3.00000E+01  5.00000E+01  0.00000E+00

NUMBER,UPPER LIMIT POINTS:=5,0,1,2,3,4.
UPPER LIMIT PROBABILITIES:=.05,.05,.05,.05
NUMBER,M/CDA POINTS:=10,0,1,2,3,4,5,6,7,8,9.
M/CDA PROBABILITIES:=.6,2,1,1,0,0,0,0,0,0.
NUMBER,OUTPUT POINTS:=10,0,1,2,3,4,5,10,20,30,50.

FIRST      LAST      PROBABILITY
0.00000E+01  1.00000E+00  7.541667E-01
1.00000E+00  2.00000E+00  1.489556E-01
2.00000E+00  3.00000E+00  5.27777E-02
3.00000E+00  4.00000E+00  2.50000E-02
4.00000E+00  5.00000E+00  8.00000E-03
5.00000E+00  1.00000E+01  0.00000E+00
1.00000E+01  2.00000E+01  0.00000E+00
2.00000E+01  3.00000E+01  0.00000E+00
3.00000E+01  5.00000E+01  0.00000E+00

NUMBER,UPPER LIMIT POINTS:=2,0,5.
UPPER LIMIT PROBABILITIES:=1.
NUMBER,M/CDA POINTS:=3,0,10.
M/CDA PROBABILITIES:=.7,0.3
NUMBER,OUTPUT POINTS:=10,0,1,2,3,4,5,10,11,12,20.

FIRST      LAST      PROBABILITY
0.00000E+01  1.00000E+00  4.307499E-01
1.00000E+00  2.00000E+00  4.307499E-01
2.00000E+00  3.00000E+00  4.415185E-02
3.00000E+00  4.00000E+00  4.415185E-02
4.00000E+00  5.00000E+00  4.415185E-02
5.00000E+00  1.00000E+01  0.00000E+00
1.00000E+01  1.10000E+01  0.00000E+00
1.10000E+01  1.20000E+01  0.00000E+00
1.20000E+01  2.00000E+01  0.00000E+00

NUMBER,UPPER LIMIT POINTS:=5,STOP
READY.

```

2.12 SCALE PROBABILITIES

A program called "HEX" was written, utilizing the interval concept, to compute an estimate of probabilities of a series of scale-related quantities. These quantities are all associated with the geometry and mass of a homogeneous spherical particle.

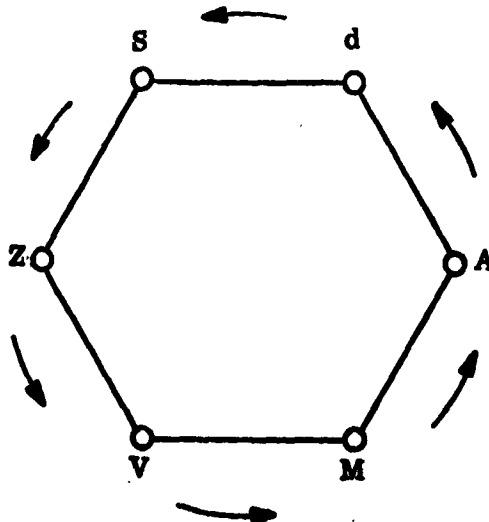
The interval concept is valid when in computing the distribution of a function of several variables, say $\xi = f(x_1, x_2, \dots, x_p)$, the intervals (or input grid) are chosen small enough so that $\frac{\partial \xi}{\partial x_j}$ ($j = 1, \dots, p$) do not change sign in the given p -dimensional regions.

The quantities under consideration are:

- d, diameter
- s, surface area
- z, ballistic parameter (M/CdA)
- V, volume
- M, mass
- A, cross - sectional area

They are all related in such a way that the above restriction is satisfied. In fact, the functions (30 in all) are all one-to-one for any given interval.

The program HEX allows one to compute the probability distribution of function 1 through 6 given any other. The 30 relations, however, are avoided and reduced to 6 by the suggestion of E. Berger. Instead, the relations are computed recursively in what may be thought of as a counterclockwise direction around the rim of a hexagon.



The six functions are given by:

1. Given d: $S = \pi d^2$
2. Given S: $Z = \frac{2\sigma}{3} \sqrt{\frac{S}{\pi}} C_D$
3. Given Z: $V = \frac{9\pi}{16} \frac{(C_D Z)^3}{\delta}$
4. Given V: $M = V\delta$
5. Given M: $A = \frac{\pi}{4} \left(\frac{\delta M}{\pi \sigma} \right)^{2/3}$
6. Given A: $d = \sqrt{\frac{4A}{\pi}}$

Where σ is the density of the particle and C_D is the (unitless) drag coefficient.

Note that the units cancel out appropriately so that it is necessary only that the given function be consistent in units with δ

For example if we start with $d = \text{cm}$, then S is cm^2 , Z is gm/cm^2 (if δ is gm/cm^3), V is cm^3 , M is gm , A is cm^2 , and d is cm .

The question may arise concerning the loss of significance encountered in "going around the horn."

This turns out to be no serious problem in the test runs encountered so far. To illustrate this, a program was written which, initially calls for δ and C_D .

Following this, it calls for the code and related functional value. The program then computes "around the horn" to the given function and prints this on the following line.

In the following runs, no round off was observed out to seven digits.

LOAD LIMITS 06031 17766
I=1.E-3, 1.E-2
I=1.0001

1.000000E-04

I=2.0001

1.000000E-04

I=3.0001

1.000000E-04

I=4.0001

1.000000E-04

I=5.0001

1.000000E-04

I=6.0001

1.000000E-04

I=1.1234567

1.234567E-01

I=2.1234567

1.234567E-01

I=3.1234567

1.234567E-01

I=4.1234567

1.234567E-01

I=5.1234567

1.234567E-01

I=6.1234567

1.234567E-01

I=88STOP
READY.

That is $\delta = .001$, $C_D = .01$
and $d = S = Z + V = M = A = .0001$
Initially

Here, all the initial
quantity values are
chosen to be = .1234567

The program usage is simple and is described briefly in the following.

The program will print the title and code initially.

The first input will be the number (integer) and functional values of the given quantity (in the standard grid format).

Following this is the set of probabilities in the standard interval concept (one less than number of end points).

Next the program calls for three A quantities: given function code (integer), density, and drag coefficient (both floating).

The function code of the desired quantity is then called for (integer).

Finally the number (integer) and functional values of the desired quantity (that is, the output grid) are called for.

The resulting probabilities (out to the last non-zero value) are printed.

The program will treat this as the input distribution for further calculations. Appropriately the code and then the related number and functional values are called for.

The recursion can be halted by giving a function code ≥ 7 . The program will then call for a new input distribution.

A sample series of runs is shown below:

The first and second test the ability to restart over for a called function code ≥ 7 .

The third and fourth demonstrate the recursion and the fifth demonstrates the ability to "recreate" the input distribution.

LOAD LIMITS 07273 16325

PROGRAM TO COMPUTE SCALE PROBABILITIES

FUNCTION CODE
1, DIAMETER
2, SURFACE AREA
3, DRAG PARAMETER
4, VOLUME
5, MASS
6, CROSS-SECTIONAL AREA

NUMBER, END POINT VALUES:=3, 1., 2., 3.

PROBABILITIES:=.1,.9

GIVEN FUNCTION CODE, DENSITY, DRAG:=1, 2., 3.

READ NEXT FUNCTION CODE:=2

NUMBER, POINTS FOR NEXT DENSITY:=5, 0., 1., 5., 10., 50., 100.

RESULTING PROBABILITIES

0.000000E-01 1.971831E-02 5.305165E-02 9.272300E-01
CHECK SUM = 1.000000

READ NEXT FUNCTION CODE:=7

NUMBER, END POINT VALUES:=3, 1., 2., 3.

PROBABILITIES:=.1,.9

FIVEN FUNCTION CODE, DENSITY, DRAG:=1, 2., 3.

READ NEXT FUNCTION CODE:=2

NUMBER, POINTS FOR NEXT DENSITY:=5, 0., 1., 5., 10., 50.

RESULTING PROBABILITIES

0.000000E-01 1.970831E-02 5.305165E-02 9.272300E-01
CHECK SUM = 1.000000

READ NEXT FUNCTION CODE:=8

NUMBER, END POINT VALUES:=5, 0., 1., 5., 10., 50.

PROBABILITIES:=.1705102, .8294898, 0., 0.

GIVEN FUNCTION CODE, DENSITY, DRAG:=3, 2., 3.

READ NEXT FUNCTION CODE:=6

NUMBER, POINTS FOR NEXT DENSITY:=10, 0., 1., 5., 10., 50., 100., 200.,
:=300., 400., 500.

RESULTING PROBABILITIES

4.233402E-02 1.365265E-01 4.346252E-02 3.477002E-01 4.294267E-01
CHECK SUM = 1.000000

READ NEXT FUNCTION CODE:=SSTOP
READY.

"REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR."

An associated program with "HEX" is the program known as "SPHERE".

The input is similar to that of "HEX", but the probabilities are not required. Only the function code, density, drag parameter and end points are needed.

The program will provide a spectrum of end points values for all the functions associated with the given end points. Of course, whatever probability is required will hold for all the end point values across each function.

PROGRAM TO COMPUTE PARTICLE PARAMETERS

FUNCTION CODE

1. DIAMETER
2. SURFACE AREA
3. DRAG PARAMETER
4. VOLUME
5. MASS
6. CROSS-SECTIONAL AREA

GIVEN FUNCTION CODE, DENSITY, DRAG:=1,2,4.

NUMBER, POINTS:=5,9,1,2,3,4.

DIAMETER	SURFACE	DRAG	VOLUME	MASS	CROSS-SEC
0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01
1.0000E+00	3.1416E+00	3.3333E-01	5.2360E-01	1.6470E+00	7.2547E-01
2.0000E+00	1.2566E+01	6.6667E-01	4.1888E+00	3.2776E+00	3.1416E+00
3.0000E+00	2.8274E+01	1.0000E+00	1.4137E+01	2.2274E+01	7.0685E+00
4.0000E+00	5.0265E+01	1.3333E+00	3.3510E+01	6.7015E+01	1.2566E+01

GIVEN FUNCTION CODE, DENSITY, DRAG:=2,2,4.

NUMBER, POINTS:=5,9,3.1416,12.566,27.274,52.065

DIAMETER	SURFACE	DRAG	VOLUME	MASS	CROSS-SEC
0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01
1.0000E+00	3.1416E+00	3.3333E-01	5.2360E-01	1.6470E+00	7.2547E-01
2.0000E+00	1.2566E+01	6.6666E-01	4.1888E+00	3.2776E+00	3.1416E+00
3.0000E+00	2.8274E+01	9.9999E-01	1.4137E+01	2.2274E+01	7.0685E+00
4.0000E+00	5.0265E+01	1.3333E+00	3.3510E+01	6.7015E+01	1.2566E+01

GIVEN FUNCTION CODE, DENSITY, DRAG:=3,2,4.

NUMBER, POINTS:=5,9,3.3333,6.6667,1,1,1.3333

DIAMETER	SURFACE	DRAG	VOLUME	MASS	CROSS-SEC
0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01
9.9999E-01	3.1415E+00	3.3333E-01	5.2359E-01	1.6470E+00	7.2547E-01
0.0000E+00	1.2566E+01	6.6667E-01	4.1888E+00	3.2776E+00	3.1416E+00
3.0000E+00	2.8274E+01	1.0000E+00	1.4137E+01	2.2274E+01	7.0685E+00
3.9999E+00	5.0265E+01	1.3333E+00	3.3510E+01	6.7015E+01	1.2566E+01

GIVEN FUNCTION CODE, DENSITY, DRAG:=4,2.,4.

NUMBER, POINTS:=5,0.,.52358,4.1889,14.137,33.508

DIAMETER	SURFACE	DRAG	VOLUME	MASS	CROSS-SEC
0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01
9.9999E-01	3.1415E+00	3.3333E-01	5.2358E-01	1.0472E+00	7.8538E-01
2.0000E+00	1.2567E+01	6.6667E-01	4.1889E+00	8.3778E+00	3.1416E+00
3.0000E+00	2.8274E+01	1.0000E+00	1.4137E+01	2.8274E+01	7.0685E+00
3.9999E+00	5.0263E+01	1.3333E+00	3.3508E+01	6.7016E+01	1.2566E+01

GIVEN FUNCTION CODE, DENSITY, DRAG:=5,2.,4.

NUMBER, POINTS:=5,0.,1.0472,8.3776,28.274,67.021

DIAMETER	SURFACE	DRAG	VOLUME	MASS	CROSS-SEC
0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01
1.0000E+00	3.1416E+00	3.3333E-01	5.2360E-01	1.0472E+00	7.8540E-01
2.0000E+00	1.2566E+01	6.6667E-01	4.1888E+00	8.3776E+00	3.1416E+00
3.0000E+00	2.8274E+01	1.0000E+00	1.4137E+01	2.8274E+01	7.0685E+00
4.0000E+00	5.0266E+01	1.3333E+00	3.3511E+01	6.7021E+01	1.2566E+01

GIVEN FUNCTION CODE, DENSITY, DRAG:=6,2.,4.

NUMBER, POINTS:=5,0.,.78540,3.1416,7.0686,12.566

DIAMETER	SURFACE	DRAG	VOLUME	MASS	CROSS-SEC
0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01	0.0000E-01
1.0000E+00	3.1416E+00	3.3333E-01	5.2360E-01	1.0472E+00	7.8540E-01
2.0000E+00	1.2566E+01	6.6667E-01	4.1888E+00	8.3776E+00	3.1416E+00
3.0000E+00	2.8274E+01	1.0000E+00	1.4137E+01	2.8274E+01	7.0686E+00
3.9999E+00	5.0264E+01	1.3333E+00	3.3509E+01	6.7018E+01	1.2566E+01

GIVEN FUNCTION CODE, DENSITY, DRAG:=\$STOP
READY.

2.13 GENERAL COMBINING OF RANDOM VARIABLES

As an aid for general engineering analysis of the probability of combinations of variates, a program called "CÖMBI" was written to accommodate such problems.

This program is a generalization of "BUGS", in that it now allows for not only addition but subtraction, multiplication and division of random variates.

The user's instructions are similar to that of "BUGS", with the only extra requirement that the binary operation code be entered.

```
PROGRAM TO COMBINE RANDOM VARIATES

OPERATION CODE
ADD, 1
SUBTRACT, 2
MULTIPLY, 3
DIVIDE, 4
RESTART, 5 OR GREATER

NUMBER,POINTS FOR FIRST DENSITY:=2,1,2,5.
FIRST SET OF PROBABILITIES:=.4,.6
NUMBER,POINTS FOR RESULTING DENSITY:=10,0,1,2,3,4,5,10.
I:=20,50,100.

READ OPERATION CODE(I-4):=1
NUMBER,POINTS FOR NEXT DENSITY:=2,1,2.
NEXT SET OF PROBABILITIES:=1.

*****RESULTING PROBABILITIES*****
0.000000E-01 0.000000E-01 2.000000E-01 3.500000E-01 1.500000E-01
3.000000E-01
CHECK SUM = 1.000000

READ OPERATION CODE(I-4):=5

NUMBER,POINTS FOR FIRST DENSITY:=3,1,2,5.
FIRST SET OF PROBABILITIES:=.4,.6
NUMBER,POINTS FOR RESULTING DENSITY:=10,0,1,2,3,4,5,10.
I:=20,50,100.

READ OPERATION CODE(I-4):=2
NUMBER,POINTS FOR NEXT DENSITY:=2,-2,-1.
NEXT SET OF PROBABILITIES:=1.

*****RESULTING PROBABILITIES*****
0.000000E-01 0.000000E-01 2.000000E-01 3.500000E-01 1.500000E-01
3.000000E-01
CHECK SUM = 1.000000
```

READ OPERATION CODE(1-4):=5

NUMBER,POINTS FOR FIRST DENSITY:=3,1.,2.,5.

FIRST SET OF PROBABILITIES:=.4,.6

NUMBER,POINTS FOR RESULTING DENSITY:=10,0.,1.,2.,3.,4.,5.,10.
:=20.,50.,100.

READ OPERATION CODE(1-4):=3

NUMBER,POINTS FOR NEXT DENSITY:=2,1.,2.

NEXT SET OF PROBABILITIES:=1.

*****RESULTING PROBABILITIES*****
0.000000E-01 1.333333E-01 2.000000E-01 2.000000E-01 7.500000E-02
3.750000E-01
CHECK SUM = 1.000000

READ OPERATION CODE(1-4):=5

NUMBER,POINTS FOR FIRST DENSITY:=3,1.,2.,5.

FIRST SET OF PROBABILITIES:=.4,.6

NUMBER,POINTS FOR RESULTING DENSITY:=10,0.,1.,2.,3.,4.,5.,10.
:=20.,50.,100.

READ OPERATION CODE(1-4):=4

NUMBER,POINTS FOR NEXT DENSITY:=2,1.,2.

NEXT SET OF PROBABILITIES:=1.

*****RESULTING PROBABILITIES*****
1.333333E-01 4.166667E-01 1.500000E-01 1.500000E-01 1.500000E-01
CHECK SUM = 1.000000

READ OPERATION CODE(1-4):=SSTOP
READY.

2.14 REFERENCES FOR SECTION 2

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2.15 ACKNOWLEDGEMENTS

This is to acknowledge the assistance of Mrs. K. Maddock in the preparation of several of the programs described in this section.

SECTION 3
AN APPROACH TO THE EVALUATION OF
THE THERMAL INACTIVATION OF MICROORGANISMS
DURING MARS ENTRY

By

M. A. Martin

TABLE OF CONTENTS

Section		Page
3	ENTRY	3-1
3.1	Introduction and Summary	3-1
3.2	Survival Ratio (\hat{i}) and Lethality Integral (I_L)	3-1
3.3	Decimal Reduction Time (D) and Thermal Death Time (F)	3-3
3.4	Curve τ (t') for Thermal Resistance of Dry Spores	3-5
3.5	Functionalization of τ (t')	3-6
3.6	Computer Programs for Determining I_L	3-8
3.7	Parameters Affecting I_L	3-9
3.8	Computer Runs for Determining I_L	3-11
3.9	Variation of I_L with Emissivity (ϵ), Intermediate Variable (ϵ_2)	3-14
3.10	Variation of ϵ_2 with Ballistic Coefficient (β)	3-16
3.11	Variation of ϵ_2 with Entry Angle (γ_E)	3-17
3.12	Variation of ϵ_2 with Entry Velocity (V_E)	3-17
3.13	Functionalization of I_L	3-18
3.14	Changes in Parameters	3-24
3.15	Organisms Carried by Nonviable Particles	3-25
3.16	References	3-26
3.17	Acknowledgements	3-26

LIST OF ILLUSTRATIONS

Figure		Page
3-1	Thermal Resistance of Dry Spores	3-5
3-2	Temperature Histories	3-12
3-3	Variation of I_L with ϵ	3-14
3-4	Variation of ϵ_2 with ϵ_1	3-15
3-5	Variation of ϵ_2 with β	3-17
3-6	Variation of ϵ_2 with γ_E	3-17
3-7	Variation of ϵ_2 with V_E	3-18
3-8	Variation of RMS with c	3-20
3-9	Evaluation of Functionalization	3-23

SECTION 3

ENTRY

3.1 INTRODUCTION AND SUMMARY

An approach to evaluate the potential thermal kill of singular bacteria or small aggregates or clumps during Mars entry has been outlined in Reference 3-1. The concept of a lethality integral (I_L) permits the calculation of the survival ratio (f) under varied conditions.

In the preliminary investigation of the method reported here, the analysis of the effects of four parameters included: the ballistic coefficient, β ; the entry angle, γE ; the entry velocity, V_E ; and the particle emissivity, ϵ . Within the range of values used for these four parameters, it has been possible to determine an algorithm to compute the lethality integral and, hence, the survival ratio (f) for the effects of any combination of these four parameters.

The development of this algorithm is explained, and the results of the functionalization of I_L are evaluated.

Changes made in the choice and the values of the parameters for further investigation of this approach, now in progress, are mentioned.

Results of calculations made on individual or singular living microbial cells or spores carried by nonviable particles are given.

3.2 SURVIVAL RATIO (f) AND LETHALITY INTEGRAL I_L

The kinetics of thermal death of microorganisms can be defined by the differential equation

$$\frac{dN}{N dt} = -K(t') \quad (3-1)$$

If N is the number of living organisms at time t , Equation 3-1, expresses that the relative rate of change of the number of living organisms is a constant depending upon the temperature t' .

Integration of Equation 3-1 yields

$$f = \frac{N}{N_0} = e^{-\int_{t_0}^t K(t') dt} \quad (3-2)$$

which gives the survival ratio f (ratio of the number N of living organisms at time t to the number N_0 of living organisms at time t_0) as a function of the time history $t'(t)$ of the temperature.

If the temperature, t' , is constant, Equation 3-2 reduces to (if $t_0 = 0$)

$$f = \frac{N}{N_0} = e^{-K(t') \cdot t} \quad (t' = \text{constant}) \quad (3-3)$$

In particular, the time $\tau(t')$ necessary to produce a specified survival ratio $f\tau$ at temperature t' is given by

$$f\tau = e^{-K(t') \tau(t')} \quad (t' = \text{constant}) \quad (3-4)$$

If the time τ is known for a specified $f\tau$, inversion of Equation 3-4 yields

$$K(t') = -\frac{\ln f\tau}{\tau(t')} \quad (3-5)$$

In all our equations, the symbol \ln represents a natural logarithm, and the symbol \log represents a decimal logarithm.

Replacing, in Equation 3-2, $K(t')$ by its value from Equation 3-5 yields

$$f = \frac{N}{N_0} = e^{\ln f\tau \int_{t_0}^t \frac{dt}{\tau(t')}} = 10^{\log f\tau \int_{t_0}^t \frac{dt}{\tau(t')}} \quad (3-6)$$

We can then define a lethality integral I_L by

$$I_L = \int_{t_0}^t \frac{dt}{\tau(t')} \quad (3-7)$$

The lethality integral, I_L , is the classical "sterility" considered in the food industry (Reference 3-2).

In our investigation, we have used:

$$f\tau = 10^{-12} \quad (3-8)$$

Hence, in our case, Equation 3-6 can be written

$$f = \frac{N}{N_0} = 10^{-12} I_L \quad (3-9)$$

For the purpose of this investigation, we have assumed that any survival ratio smaller than 10^{-4} is considered as meeting the planetary quarantine requirement. Consequently, the range of values of interest for I_L is from 0 to 1/3.

3.3 DECIMAL REDUCTION TIME (D) AND THERMAL DEATH TIME (F)

Equation 3-3 can be written

$$f = \frac{N}{N_0} = 10^{-K'(t') t} \quad (t' = \text{constant}) \quad (3-10)$$

with

$$K'(t') = (\log e) \cdot K(t') = 0.43429 K(t') \quad (3-11)$$

The constant $K'(t')$ is determined experimentally from the measurement of the survival ratio, f , for a known time, t , at the specified temperature, t' . Instead of $K'(t')$, the biologists use its reciprocal $D(t')$, hence Equation 3-10 can be written

$$f = \frac{N}{N_0} = 10^{-\frac{t}{D(t')}} \quad (t' = \text{constant}) \quad (3-12)$$

If the time, t , is equal to $D(t')$, the survival ratio, f , is $1/10$; that is, $D(t')$ represents the time to reduce the number of viable organisms in a population to one tenth of its initial value, hence the term Decimal reduction time given to D .

When $D(t')$ is known, the time τ necessary to produce a specified survival ratio $f\tau$ is given by

$$f\tau = 10^{-\frac{\tau}{D(t')}} \quad (t' = \text{constant}) \quad (3-13)$$

hence

$$\tau = D(t') \cdot \left[-\log f\tau \right] \quad (3-14)$$

If N_0 is the initial population and $N\tau$ the population at time τ , we have

$$-\log f\tau = -\log \frac{N\tau}{N_0} = \log N_0 - \log N\tau \quad (3-15)$$

Hence

$$\tau = D(t') \left[\log N_0 - \log N\tau \right] \quad (3-16)$$

$N\tau$ may be interpreted as the probability of having, at time τ , one living organism out of an initial population N_0 maintained at constant temperature t .

The F value (time to sterilize, more commonly referred to as thermal death time) familiar to the biologists, is derived from the D value by the equation of Schmidt (Reference 3-3), which can be written, with our notations,

$$F = D(t') \left[\log N_0 + 1 \right] \quad (3-17)$$

or by Hobby's modification of Schmidt's Equation (Reference 4)

$$F = D(t') \left[\log N_0 + a\tau - \log N_T \right] \quad (3-18)$$

Equation 3-18 differs from Equation 3-16 only by the term $a\tau$. Hobby takes $a\tau$ equal to 2; Koesterer used the value 1 for $a\tau$. Actually when N_T is specified, the value 0 should be used for this term. Adding it is equivalent to replacing N_T by

$$N_T' = 10^{-a\tau} \cdot N_T \quad (3-19)$$

3.4 CURVE $\tau(t')$ FOR THERMAL RESISTANCE OF DRY SPORES

From D values obtained at temperatures of 80°C, 100°C to 150°C by 5°C increments, and 160°C, (Reference 3-5), and from D values obtained from Decker's work for higher temperature (Reference 3-6), M. Koesterer established a curve (Figure 3-1) of τ as a function of the temperature t' for $N_0 = 10^8$ and $N_T = 10^{-4}$, that is, for $f\tau = 10^{-12}$ as mentioned in Equation 3-8.

Figure 3-1 is actually a curve of F values. Koesterer used the value 1 for $a\tau$; hence, the curve really corresponds to a value 10^{-13} for $f\tau$.

This fact was discovered only recently, and since the purpose of this preliminary investigation was to develop a method for the functionalization of the lethality integral, I_L , the value 12 has been retained for this report.

Correction would involve replacing 12 by 13

in Equation 3-9 and in all the calculations which convert I_L into values of survival ratios.

The use of 12 rather than 13 provides conservative estimates for the survival ratio.

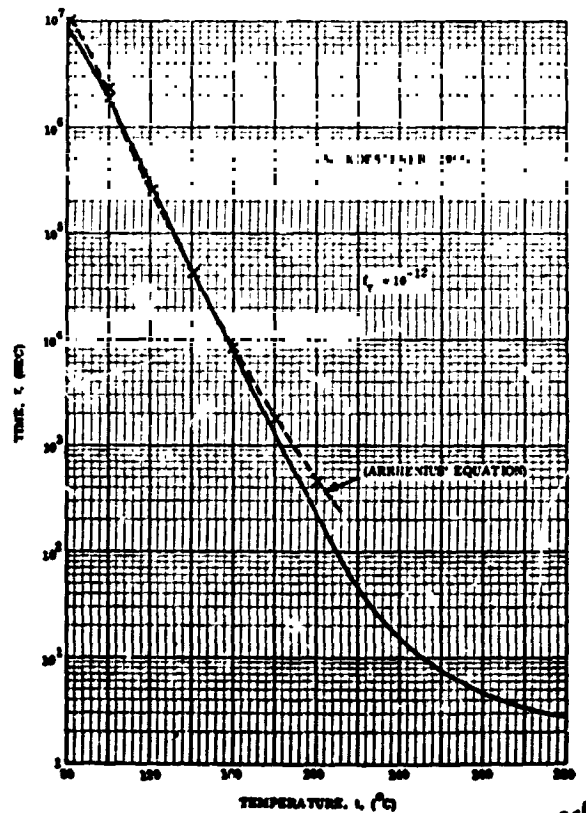


Figure 3-1. Thermal Resistance of Dry Spores

Arrhenius-Van t'Hoff's theory provides a theoretical expression for $D(t')$:

$$D(t') = A e^{-\frac{E_{dh}}{RT}} \quad (3-20)$$

In Equation 3-20, A is a constant referred to as frequency factor, R is the gas constant, E_{dh} is the thermal inactivation energy, and T is the absolute temperature (Reference 3-7).

We can transform Equation 3-20 into

$$\log D(t') = \frac{a_D}{T} + b_D \quad (3-21)$$

The coefficients a_D and b_D were determined to match Koesterer's curve for the temperatures 100°C and 160°C . The resulting equation was

$$\log D(t') = \frac{6355.1}{T} - 11.826 \quad (3-22)$$

Since, in our case, τ is equal to 12 times the D value, values of τ were computed with Equation 3-22 for values of t' from 80°C to 210°C . The results are shown in Figure 3-1 (dotted line).

It can be seen that for high temperatures, the time required to produce a specified survival ratio is less than that predicted by the kinetic theory.

3.5 FUNCTIONALIZATION OF $\tau(t')$

For the purpose of our investigation, Koesterer's curve (Figure 3-1) was assumed to represent actual values.

Figure 3-1 represents $\log \tau$ as a function of the temperature t' . For t' between 80°C and 210°C , the curve is a straight line. The portion of the curve for t' between 210°C and 320°C was approximated by a portion of a rectangular hyperbola.

Specifically, the following functions have been used:

$$\tau = e^{(-0.089810 t' + 23.443)} \quad \text{for } t' \leq 210^{\circ}\text{C} \quad (3-23)$$

$$\tau = e^{\frac{-0.797776 t' + 416.679}{t' - 155.656}} \quad \text{for } t' > 210^{\circ}\text{C} \quad (3-24)$$

In our calculations, Equation 3-24 was also applied to temperatures higher than the maximum temperature (320°C) for which experimental data exist. The reason is that Equation 3-24 still provides reasonable extrapolated values for these higher temperatures. Furthermore, at 320°C, an exposure time of 0.89 second is sufficient to produce a survival ratio of 10^{-4} ; hence, when that temperature is reached or exceeded during Mars entry, the thermal kill is almost instantaneous. The exact value of n in the survival ratio 10^{-n} cannot be computed accurately, but is not pertinent when n is larger than 4.

Table 3-1 provides a synopsis of the quality of the functionalization of τ . Between 100°C and 210°C, the % error should be theoretically zero, since Koesterer's curve is a straight line and Equation 3-23 represents also a straight line when τ is plotted in logarithmic scale; the small errors are due to errors in interpreting the curve.

Table 3-1 shows that for the range of experimental temperatures, the value of τ functionalized by Equation 3-23 or 3-24 does not differ from the observed value by more than a few percent.

Table 3-1. Functionalization of τ

t' (°C)	τ (seconds)		Percentage of Error	t' (°C)	τ (seconds)		Percentage of Error
	Read	Calculated			Read	Calculated	
100	1 920 000.	1 910 000.	-0.5	260	7.5	7.43	-0.9
120	313 000.	317 000.	1.3	270	6.0	5.81	-3.2
140	51 000.	52 500.	2.9	280	1.8	4.73	-1.5
160	8 400.	8 200.	3.8	290	4.0	3.97	-0.7
180	1 390.	1 450.	4.3	300	3.4	3.42	0.6
200	235.	240.	2.1	310	3.0	3.00	0.0
210	100.	98.0	-2.0	320	2.66	2.67	0.4
220	43.5	42.4	-2.5	340		2.20	
230	23.0	23.0	0.0	360		1.88	
240	14.3	14.4	0.7	380		1.66	
250	10.0	10.0	0.0	400		1.49	

3.6 COMPUTER PROGRAMS FOR DETERMINING I_L

The method of determination of a particle temperature during an entry trajectory is described in Reference 3-8.

A trajectory program is first run with selected initial conditions to provide values of the free molecular heat flux \dot{q}_{FM} (in Btu/ft²-sec) as functions of time. At each instant, \dot{q}_{FM} is proportional to the atmospheric density ρ_a at the particle position and to the cube of the particle velocity

$$\dot{q}_{FM} \propto \rho_a V^3 \quad (3-25)$$

The heat balance equation can be written (for a sphere)

$$\dot{q}_{FM} + \alpha \bar{S} = K_1 \epsilon \sigma T^4 + K_2 r \rho C_P \frac{dT}{dt} \quad (3-26)$$

In Equation 3-26,

α is the solar absorptivity of the particle

\bar{S} is the solar constant for Mars (a value of 0.0653 Btu/ft²-sec was used)

ϵ is the particle emissivity

σ is the Stefan-Boltzmann constant (4.76×10^{-13} Btu/ft-sec-⁰R)

r is the particle radius (ft)

ρ is the particle density (lb/ft³)

C_P is the specific heat capacity of the particle (Btu/lb-⁰R)

T is the absolute temperature of the particle (⁰R)

K_1 has value 4 for a spherical particle, and π for a cylindrical particle

K_2 has value 4/3 for a spherical particle, and $\pi/2$ for a cylindrical particle

The input to the Thermodynamics program includes the values of \dot{q}_{FM} for all the times needed by the computer program and the value T_0 of T at the initial time t_0 . At each time t_n , the derivative $\frac{dT}{dt}$ is calculated by Equation 3-26, and the value T for the next time is obtained by integration of this derivative.

A subroutine has been added to the thermodynamics program to compute I_L as follows: at each time t_n , the absolute temperature T_n of the particle (in $^{\circ}R$) is converted to a value t'_n in $^{\circ}C$ from which $\frac{1}{\tau_n}$ is computed from Equation 3-23 or 3-24, by simple change of sign in the exponent. If $I_{L(n-1)}$ is the value of the integral I_L up to the preceding time t_{n-1} , the corresponding value of τ_n , the value I_{Ln} of I_L at time t_n is computed, according to the trapezoidal rule of integration, by

$$I_{Ln} = I_{L(n-1)} + \left(\frac{t_n - t_{n-1}}{2} \right) \left(\frac{1}{\tau_n} + \frac{1}{\tau_{n-1}} \right) \quad (3-27)$$

The value of I_{Ln} for the last value of t_n processed by the thermodynamics program represents the lethality integral I_L .

3.7 PARAMETERS AFFECTING I_L

In the preliminary investigation reported here, the particles were assumed to be spherical. It is shown in Reference 3-8 that the temperatures obtained with cylindrical particles, with the end effects neglected, are slightly higher than those obtained with spherical particles and hence produce slightly higher lethality integrals. Consideration of spherical particles is then favorable to survival of the particle.

The particles have been assumed to have a temperature of 500 $^{\circ}R$ at the start of the entry trajectory, the altitude h_0 of which has been maintained at the constant value of 721,000 feet (entry altitude).

The VM3 atmosphere has been selected because it is less dense than the VM8 atmosphere and hence provides conservative estimates for particle survival.

The solar absorptivity α has been maintained to 1 (that is, the particle has been assumed to be in daytime entry and to absorb all the solar energy it receives).

The particle specific heat capacity C_P has been maintained equal to 0.2 Btu/lb-^oR. A constant drag coefficient c_D equal to 2 has been used.

The particles have been assumed to have a constant density equal to 68.6 lb/ft³ or 2.132 slugs/ft³.

We have varied only four parameters:

a. The ballistic coefficient

$$\beta = \frac{M}{c_D A} \quad (3-28)$$

where M is the particle mass (in slugs) and A the area (in ft²) of the particle section. The ballistic coefficient β is related to the particle radius (in feet) by

$$\beta = \frac{M}{c_D A} = \frac{\frac{4}{3} \pi r^3 \rho}{c_D \cdot \pi r^2} = \frac{4}{3} \frac{\rho}{c_D} r \quad (3-29)$$

hence

$$r = \frac{3}{4} \frac{c_D}{\rho} \beta \quad (3-30)$$

Since c_D and ρ have been maintained constant, r was determined by the value of β . Specifically, the three sets of values of β and r we have used are:

β	4.0×10^{-5}	2.2×10^{-4}	4.0×10^{-4}	slugs/ft ²
r	2.81×10^{-5}	1.55×10^{-4}	2.81×10^{-4}	feet

These values are in agreement with those of Table 4-1 of Reference 3-8.

- b. Five values of the entry angle γ_E (angle of the trajectory with the local horizontal at entry, counted positive downward) have been used:

5 degrees, 10 degrees, 20 degrees, 45 degrees, 90 degrees
(downward vertical)

- c. Five values of the entry velocity, V_E have been used:

11306, 15000, 19000, 22500, 26000 ft/sec

- d. Nine values of the particle emissivity ϵ have been used:

0.1 to 0.9 by 0.1 increment

The lethality integral I_L is a monotonic function of some of these parameters. Specifically:

I_L increases when the entry temperature T_0 increases
 I_L increases when the solar absorptivity α increases
 I_L increases when the ballistic coefficient β increases
 I_L increases when the entry velocity, V_E increases
 I_L decreases when the emissivity ϵ increases
 I_L decreases when the specific heat capacity (C_p) increases

The variation of I_L with the entry angle γ_E could not be predicted. Effectively, as shown in Figure 3-2, when γ_E increases, the maximum temperature increases, but the duration of the temperature history which significantly contributes to I_L decreases.

3.8 COMPUTER RUNS FOR DETERMINING I_L

A total of 75 trajectories was required to represent all the possible combinations of three values of β , five values of γ_E , and five values of V_E . If all the nine possible values of emissivity, ϵ , had been used for each trajectory, 675 computer runs would have been necessary. The resulting computer time and manpower necessary to prepare all the input for the computer runs would have been prohibitively high. Furthermore, a large number of values of I_L would have been outside the range of interest.

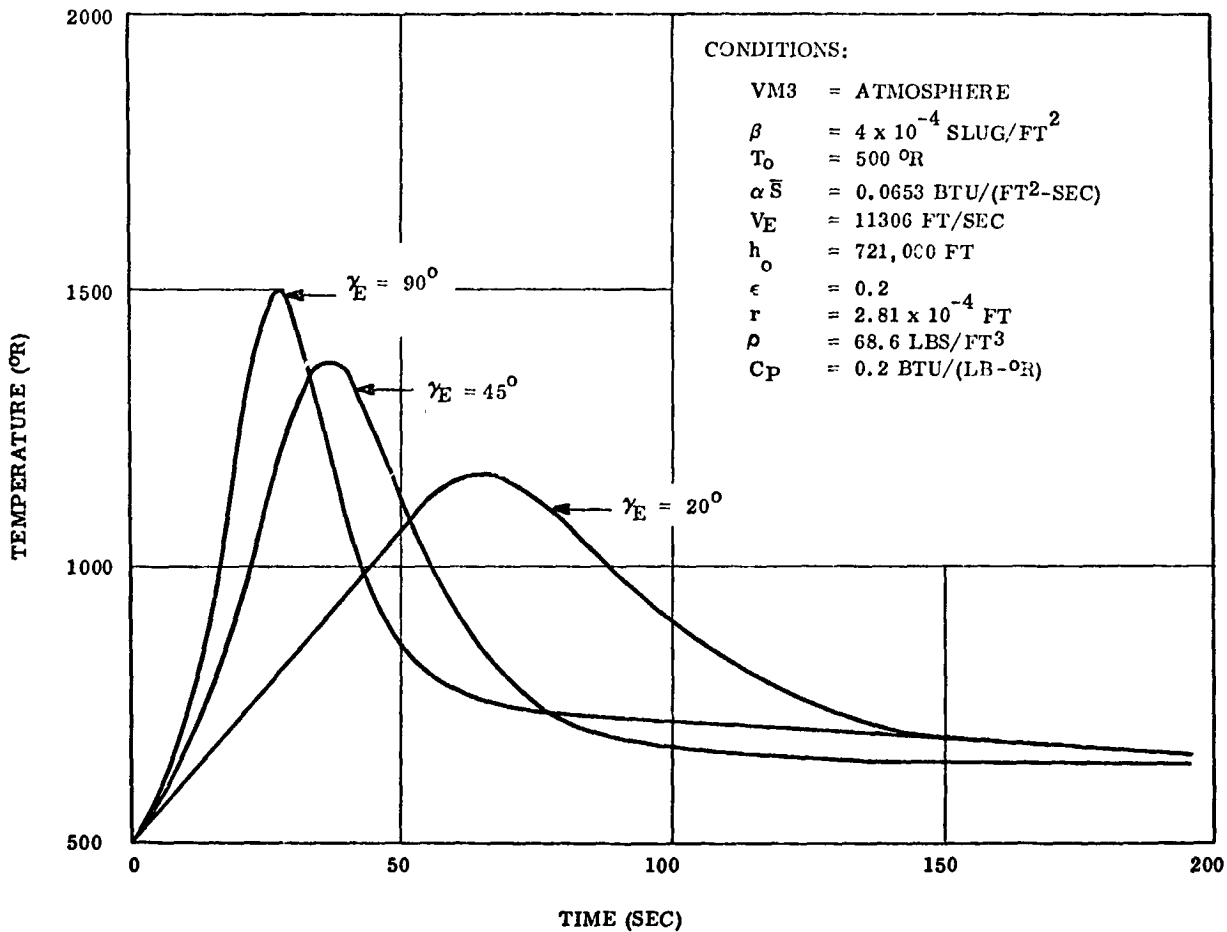


Figure 3-2. Temperature Histories

Consequently, computer runs were specified by small batches of 10 to 16. No batch was specified until the results of the preceding batch had been obtained and analyzed. In that manner, more educated guesses could be made as additional results became available.

Table 3-2 shows a synopsis of all the computer runs which were made and for which the lethality integral I_L was calculated.

A total of less than 150 computer runs was made, that is, about one-fifth of the number 675 of possible combinations of the four variables.

3.9 VARIATION OF I_L WITH EMISSIVITY ϵ . INTERMEDIATE VARIABLE ϵ_2

As it can be shown in Table 3-2, I_L varies quite nonlinearly with ϵ . It was then natural to plot $\log I_L$ as function of I_L .

Figure 3-3 shows a few of the curves which were plotted for $\beta = 2.2 \times 10^{-4}$ slug/ft². It can be seen that for I_L between 0.01 and 0.3 (range close to the range of interest 0 to 0.333), the various curves can be approximated by straight lines.

We could then define these straight lines by two parameters. We selected the value ϵ_1 of the emissivity for I_L equal to 0.3 and the value ϵ_2 of the emissivity for I_L equal to 0.01.

In order to determine the envelope of these straight lines, ϵ_2 was plotted as function of ϵ_1 . Figure 3-4 shows the result: a straight line passing through the origin.

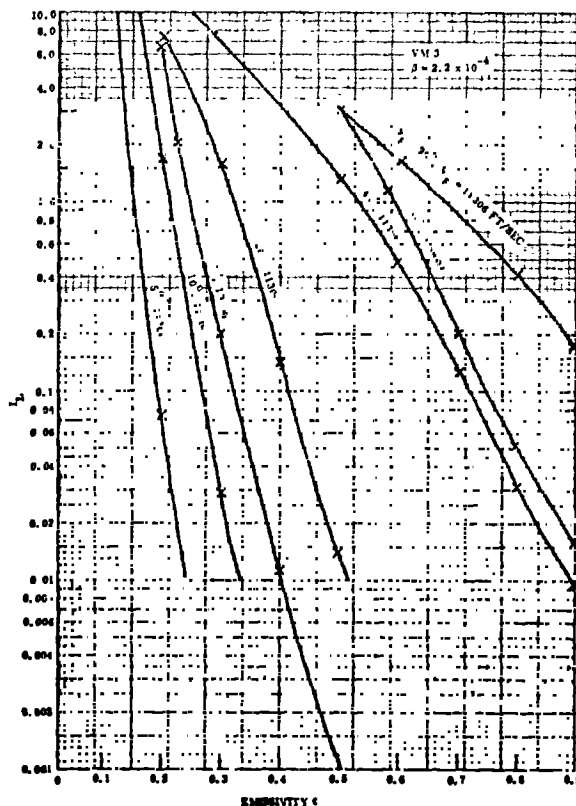


Figure 3-3. Variation of I_L with ϵ

This indicated that all the straight lines passed through a point of the $\log I_L$ axis, that is, having for coordinates in Figure 3-3.

$$\epsilon = 0 \quad \log I_{LO} = c \quad (3-31)$$

Effectively, the equation of any of the straight lines is

$$\log I_L = a \epsilon + c \quad (3-32)$$

Expressing that the line goes through the points $(\epsilon_1, \log 0.3)$ and $(\epsilon_2, \log 0.01)$ yields the equations

$$-0.52288 = \log 0.3 = a \epsilon_1 + c \quad (3-33)$$

$$-2. = \log 0.01 = a \epsilon_2 + c \quad (3-34)$$

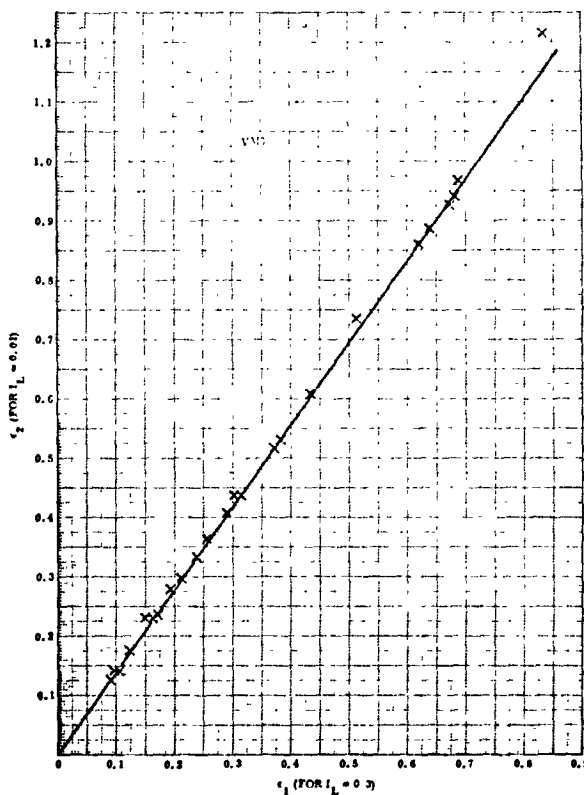


Figure 3-4. Variation of ϵ_2 with ϵ_1

From Figure 3-4, we obtained

$$\epsilon_2 = 1.38235 \epsilon_1 \quad (3-35)$$

Hence

$$a \epsilon_1 = -0.52288 - c \quad (3-36)$$

$$a \epsilon_2 = -2 - c \quad (3-37)$$

and

$$1.38235 = \frac{\epsilon_2}{\epsilon_1} = \frac{a \epsilon_2}{a \epsilon_1} = \frac{-2 - c}{-0.52288 - c} \quad (3-38)$$

Solution of Equation 3-38 yields

$$c = 3.340 = \text{constant} \quad (3-39)$$

Since all the straight lines of Figure 3-4 passed through the same point defined by Equations 3-31 and 3-39, each straight line could be defined by a single parameter. We selected ϵ_2 .

Elimination of a between Equations 3-32 and 3-34 yields

$$\log I_L = c - \frac{(c+2)\epsilon}{\epsilon_2} \quad (3-40)$$

or

$$\epsilon_2 = \frac{(c+2)\epsilon}{c - \log I_L} \quad (3-41)$$

For each data point (combination β , γ_E , V_E , ϵ , and corresponding I_L obtained by computer run) it is possible to compute ϵ_2 by Equation 3-41.

The problem was then to express ϵ_2 as function of the three remaining variables β , γ_E and V_E .

The maximum value of ϵ_2 of interest is obtained for $\epsilon = 0.9$ and $I_L = 1/3$ and is 1.244 with the value of Equation 3-39 used for c .

3.10 VARIATION OF ϵ_2 WITH BALLISTIC COEFFICIENT β

Figure 3-5 shows a few of the plots of ϵ_2 as function of β . (Figure 3-5 is for $V_E = 11306$ ft/sec only). It was found that every time we had data for the three values of β and a value of ϵ_2 not exceeding 1.244, the three points belonging to the case combination (γ_E , V_E) were on a straight line.

In other words, within the range of values for β that we have used, ϵ_2 was a linear function of β .

3.11 VARIATION OF ϵ_2 WITH ENTRY ANGLE (γ_E)

It was found that ϵ_2 was not a linear function of γ_E , but was a linear function of $(90^\circ - \gamma_E)^2$. This can be shown for instance, in Figure 3-6, which represents curves obtained with $\beta = 4 \times 10^{-5}$ slug/ft².

This quadratic form of the function, with symmetry with respect to 90 degrees, can easily be understood: the entry angles $90 - \alpha$ and $90 + \alpha$ define two trajectories symmetrical with respect to the downward vertical.

3.12 VARIATION OF ϵ_2 WITH ENTRY VELOCITY (V_E)

It was found that ϵ_2 could be represented by a linear function of the cube of the entry velocity V_E . This can be shown, for instance, in Figure 3-7, which represents curves obtained with $\beta = 4 \times 10^{-5}$ slug/ft².

It seems interesting to compare this dependence of I_L on V_E^3 with the dependence of \dot{q}_{FM} on V^3 as shown in Equation 3-25.

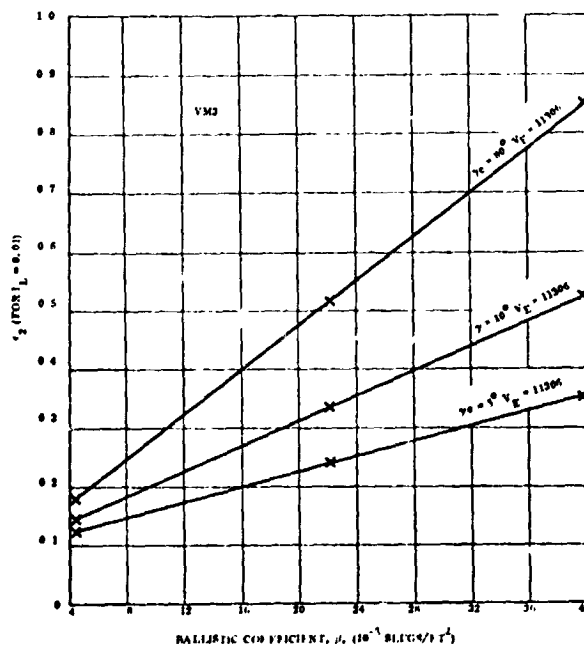


Figure 3-5. Variation of ϵ_2 with β

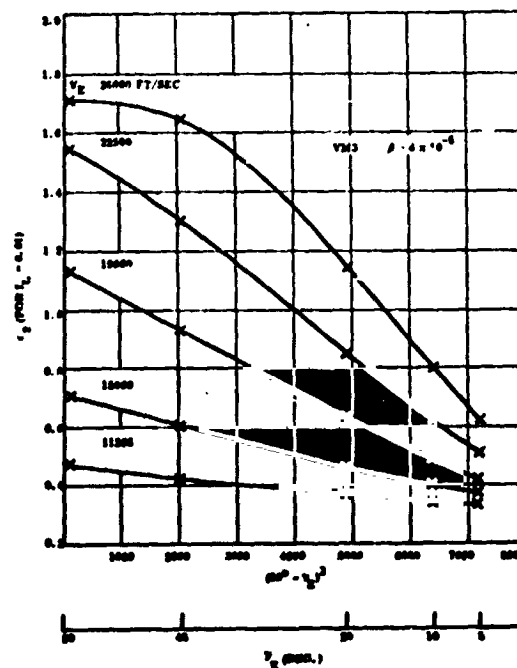


Figure 3-6. Variation of ϵ_2 with γ_E

3.13 FUNCTIONALIZATION OF I_L

For convenience, the functions β , $(90 - \gamma_E)^2$, and V_E^5 have been scaled and replaced by

$$\bar{\beta} = 10^4 \beta \quad \bar{\gamma} = \left(\frac{90 - \gamma_E}{100} \right)^2$$

$$\bar{V} = \left(\frac{V_E}{10^4} \right)^{33} \quad (3-42)$$

Since ϵ_2 is a linear function of $\bar{\beta}$, $\bar{\gamma}$, \bar{V} , it can be represented by a sum of terms of the form

$$\bar{\beta}^i \bar{\gamma}^j \bar{V}^k$$

in which the exponents i , j , k can take

values 0 or 1. Hence ϵ_2 can be approximated by a sum ϵ_{2c} of eight such terms. Specifically

$$\epsilon_2 \cong \epsilon_{2c} = a_1 + a_2 \bar{\beta} + a_3 \bar{\gamma} + a_4 \bar{V} + a_5 \bar{\beta} \bar{\gamma}$$

$$+ a_6 \bar{\gamma} \bar{V} + a_7 \bar{V} \bar{\beta} + a_8 \bar{\beta} \bar{\gamma} \bar{V} \quad (3-43)$$

For each data point $\bar{\beta}$, $\bar{\gamma}$, \bar{V} can be calculated by Equation 3-42 and ϵ_2 by Equation 3-41. We can then obtain the coefficients of the linear combination (3-43) of known functions, for instance, by least square fit.

We have assigned a weight (1 or 0) to each data point. This has permitted us to make basically all the same calculations on all data points, but to eliminate from the least square fit the data points which did not agree closely enough with the calculated fit. In that manner, we processed 119 data points, but considered only 63 data points for determining the least square fit. The other 56 points had values of I_L larger than 0.333 and hence corresponded

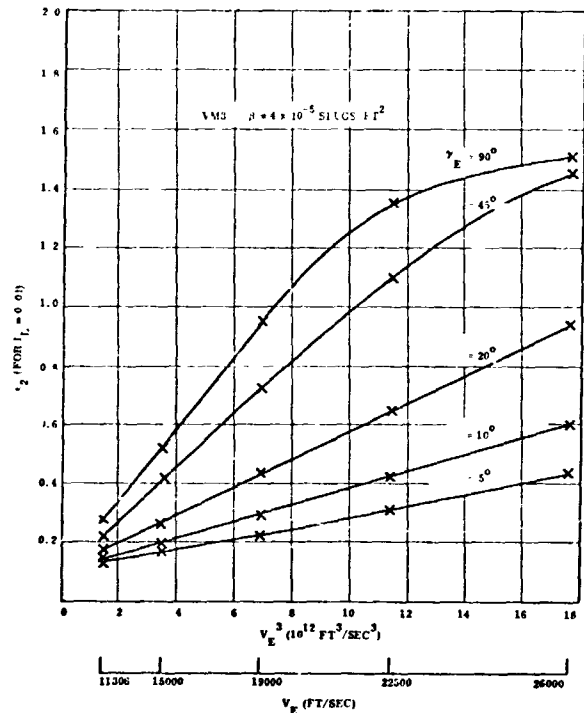


Figure 3-7. Variation of ϵ_2 With V_E

104

to complete kill (survival ratios f smaller than 10^{-4}). For these values of f smaller than 10^{-4} , the exact value of f was of no interest, as long as it was smaller than 10^{-4} . We gave a weight of zero to only 3 of the basic 63 data points.

An iterative process, operating on the value c , was used to improve the accuracy of the fit. The procedure can be explained as follows:

For a given value of c (the first value was 3.340), ϵ_2 can be computed for each data point by Equation 3-41 and the values $\bar{\beta}$, $\bar{\gamma}$, \bar{V} by (3-42). After all the data points have been processed, the coefficients a_1 to a_8 of Equation 3-43 are obtained by the classical weighted least square method. Then for each data point a computed value ϵ_{2c} is obtained by Equation 3-43, a corresponding computed lethality integral I_{LC} is obtained, according to Equation 3-40 by

$$\log I_{LC} = c - \frac{(c+2)\epsilon}{\epsilon_{2c}} \quad (3-44)$$

then

$$I_{LC} = 10^{\log I_{LC}} \quad (3-45)$$

The corresponding computed survival ratio f_c is obtained according to Equation 3-9 by

$$f_c = 10^{-12 I_{LC}} \quad (3-46)$$

For each data point, the residual

$$\Delta I_L = I_L - I_{LC} \quad (3-47)$$

difference between the true value I_L and the computed value I_{LC} is calculated. The ratio between the true survival ratio (f) and the computed survival ratio (f_c) is related to this residual by

$$\frac{f}{f_c} = 10^{-12 \Delta I_L} \quad (3-48)$$

A weighted root mean square value of the residual ΔI_L is obtained for all the N data points by

$$\text{RMS} = \sqrt{\frac{\sum_{n=1}^N W_n \Delta I_L^2}{\sum_{n=1}^N W_n}} \quad (3-49)$$

W_n being the weight assigned to the data point of sequential order (n).

The process is repeated for various values of (c) until, by trial and error, a practical minimum value is obtained for RMS. Figure 3-8 summarizes the results.

The minimum RMS was obtained with

$$c = 2.87 \quad (3-50)$$

and had for value

$$\text{RMS} = 0.02047$$

corresponding to a "root mean square" ratio $\frac{f}{f_c}$ equal to 1.75.

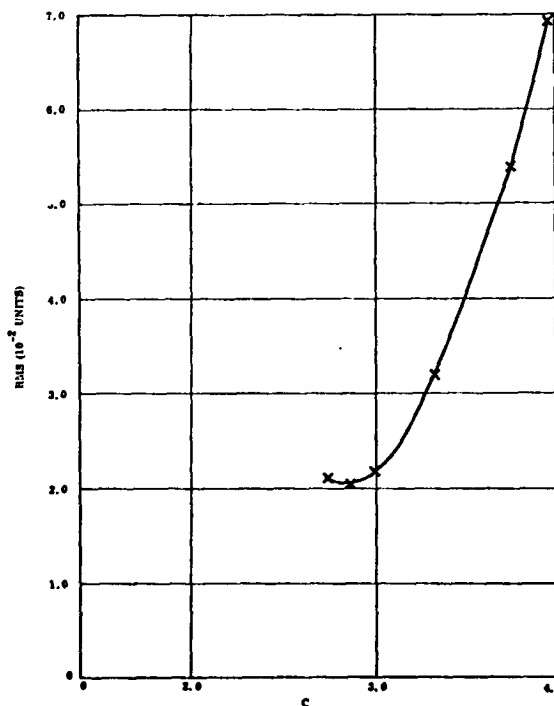


Figure 3-8. Variation of RMS With c

Table 3-3 shows the results of the calculations on each of the 119 data points with the value 2.87 for c. In that table the first eight numbers are the coefficients a_1 to a_8 of Equation 3-43. The last line shows the numerator and denominator of the fraction in Equation 3-49 and the RMS.

Table 3-3. Functionalization of I_L for c = 2.87

I	W	β	Y _E	V _E	ε	ε ₂	ε _{2c}	I _L	ΔI _L
1	0.000004	5.00	11306.	0.10	0.1343	0.1199	0.1750	0.0642	0.1108
2	0.000004	5.00	15000.	0.20	0.1711	0.1543	0.0015	0.0054	0.0011
3	0.000004	5.00	19000.	0.30	0.2202	0.2165	0.0400	0.0235	0.0165
4	0.000004	5.00	22500.	0.30	0.3091	0.2974	0.0139	0.0091	0.0048
5	0.000004	5.00	26000.	0.40	0.4283	0.4078	0.0210	0.0124	0.0086
6	0.000004	5.00	26000.	0.50	0.4372	0.4078	0.0026	0.0008	0.0012
7	0.000004	10.00	11306.	0.20	0.1238	0.1415	0.0000	0.0001	-0.0001
8	0.000004	10.00	15000.	0.20	0.1901	0.1977	0.0000	0.0000	0.0000
9	0.000004	10.00	19000.	0.30	0.2941	0.2933	0.0000	0.0000	0.0000
10	0.000004	10.00	19000.	0.40	0.3319	0.2993	0.0010	0.0002	0.0008
11	0.000004	10.00	22500.	0.30	0.4391	0.4315	0.3490	0.0000	0.0000
12	0.000004	10.00	22500.	0.40	0.4201	0.4315	0.6171	0.0000	0.0000
13	0.000004	10.00	26000.	0.30	0.5031	0.6115	0.0000	0.0000	0.0000
14	0.000004	10.00	26000.	0.40	0.6121	0.6115	0.0000	0.0000	0.0000
15	0.000004	10.00	11306.	0.20	0.1745	0.1547	0.1115	0.0000	0.0000
16	0.000004	10.00	15000.	0.20	0.2682	0.2767	0.1750	0.0000	0.0000
17	0.000004	10.00	19000.	0.40	0.4389	0.4499	0.0277	0.0000	0.0000
18	0.000004	10.00	19000.	0.70	0.4332	0.4499	0.0000	0.0000	0.0000
19	0.000004	20.00	22500.	0.50	0.6714	0.6752	0.1750	0.0000	0.0000
20	0.000004	20.00	26000.	0.70	0.9733	0.9828	0.0000	0.0000	0.0000
21	0.000004	20.00	26000.	0.70	0.9733	0.9828	0.2337	0.0000	0.0000
22	0.000004	20.00	26000.	0.80	0.9504	0.9828	0.0000	0.0000	0.0000
23	0.000004	20.00	26000.	0.90	0.9447	0.9828	0.0170	0.0000	0.0000
24	0.000004	45.00	11306.	0.20	0.2285	0.2560	0.0410	0.0000	0.0000
25	0.000004	45.00	15000.	0.30	0.4195	0.4280	0.0410	0.0000	0.0000
26	0.000004	45.00	19000.	0.60	0.7413	0.7386	0.2440	0.0000	0.0000
27	0.000004	45.00	19000.	0.70	0.7348	0.7386	0.0170	0.0000	0.0000
28	0.000004	45.00	19000.	0.90	0.7467	0.7386	0.0000	0.0000	0.0000
29	0.000004	45.00	22500.	0.80	1.1425	1.1425	0.2890	0.0000	0.0000
30	0.000004	90.00	11306.	0.20	0.2858	0.3096	0.2900	0.0000	0.0000
31	0.000004	90.00	11306.	0.30	0.2773	0.3096	0.0000	0.0000	0.0000
32	0.000004	90.00	15000.	0.40	0.5322	0.5340	0.1120	0.0000	0.0000
33	0.000004	90.00	19000.	0.70	0.9701	0.9419	0.2270	0.0000	0.0000
34	0.000004	90.00	19000.	0.70	0.9752	0.9419	0.0000	0.0000	0.0000
35	0.000004	90.00	19000.	0.90	0.9661	0.9419	0.0000	0.0000	0.0000
36	0.000004	135.00	11306.	0.20	0.2435	0.2335	0.0740	0.0000	0.0000
37	0.000004	135.00	15000.	0.30	0.3033	0.2942	0.0113	0.0000	0.0000
38	0.000004	135.00	19000.	0.30	0.4116	0.4034	0.2091	0.0000	0.0000
39	0.000004	135.00	19000.	0.40	0.4034	0.4034	0.0110	0.0000	0.0000
40	0.000004	135.00	19000.	0.50	0.4145	0.4034	0.0000	0.0000	0.0000
41	0.000004	135.00	22500.	0.20	0.5503	0.5454	0.2140	0.0000	0.0000
42	0.000004	135.00	22500.	0.30	0.3326	0.3379	0.0000	0.0000	0.0000
43	0.000004	135.00	26000.	0.40	0.5675	0.5540	0.2730	0.0000	0.0000
44	0.000004	135.00	26000.	0.70	0.9341	0.9405	0.2020	0.0000	0.0000
45	0.000004	135.00	26000.	0.70	0.9341	0.9405	0.0000	0.0000	0.0000
46	0.000004	135.00	26000.	0.90	0.9338	0.9405	0.0000	0.0000	0.0000
47	0.000004	270.00	11306.	0.20	0.3227	0.3329	0.1390	0.0000	0.0000
48	0.000004	270.00	15000.	0.70	1.0291	1.0263	0.3610	0.0000	0.0000
49	0.000004	270.00	19000.	0.50	0.5155	0.5329	0.0000	0.0000	0.0000
50	0.000004	270.00	19000.	0.70	0.9227	0.9429	0.0000	0.0000	0.0000
51	0.000004	270.00	19000.	0.90	0.9069	0.9029	0.0000	0.0000	0.0000
52	0.000004	270.00	22500.	0.40	0.5816	0.5634	0.0000	0.0000	0.0000
53	0.000004	270.00	22500.	0.20	0.3691	0.3634	0.1720	0.0000	0.0000
54	0.000004	270.00	26000.	0.30	0.3691	0.3477	0.0650	0.0000	0.0000
55	0.000004	270.00	26000.	0.40	0.3698	0.3477	0.0000	0.0000	0.0000
56	0.000004	270.00	26000.	0.50	0.3323	0.3477	0.0000	0.0000	0.0000
57	0.000004	270.00	26000.	0.70	0.5303	0.5344	0.0000	0.0000	0.0000
58	0.000004	270.00	26000.	0.40	0.5303	0.5344	0.1790	0.0000	0.0000
59	0.000004	270.00	26000.	0.50	0.5277	0.5344	0.0000	0.0000	0.0000
60	0.000004	270.00	26000.	0.70	0.5277	0.5344	0.0000	0.0000	0.0000
61	0.000004	270.00	26000.	0.70	0.5277	0.5344	0.1160	0.0000	0.0000
62	0.000004	270.00	26000.	0.70	0.5277	0.5344	0.0000	0.0000	0.0000

FOLDOUT FRAME

NOTES

ENTRANCE FRAME

FOLDOUT FRAME

2

3-21/22

107

54	0.00040	5.00	11306.	0.30	9.3691	0.3477	0.0650	0.0005	0.0185
55	0.00040	5.00	11306.	0.40	0.3698	0.3477	0.0640	0.0018	0.0022
56	0.00040	5.00	13060.	0.50	0.3333	0.5641	0.1340	0.3197	-1.1007
57	0.00040	5.00	15000.	0.60	0.3000	0.5641	0.0679	0.0242	-0.1163
58	0.00040	10.00	11306.	0.40	0.3698	0.5354	0.1790	0.1785	-0.1005
59	0.00040	10.00	11306.	0.50	0.3277	0.5354	0.0180	0.0222	-0.0042
60	0.00040	10.00	11306.	0.70	0.3507	0.5354	0.0010	0.0003	0.0057
61	0.00040	15.00	15000.	0.70	0.3507	0.9103	0.1160	0.1333	-0.0173
62	0.00040	15.00	11306.	0.70	0.3507	0.5354	0.0070	0.1044	-0.1177
63	0.00040	20.00	11306.	0.90	0.3723	0.5354	0.0075	0.0083	-0.0013
64	0.00040	5.00	26000.	0.30	0.4497	0.4675	0.4150	0.1937	-0.2243
65	0.00040	10.00	11306.	0.10	0.509	0.1415	0.4390	0.2678	0.1712
66	0.00040	20.00	19000.	0.20	0.4619	0.4239	0.5040	0.4196	0.0544
67	0.00040	45.00	19000.	0.50	0.7352	0.7356	0.3750	0.3742	0.0006
68	0.00022	45.00	11306.	0.50	0.9197	0.9229	0.4700	0.4302	0.0395
69	0.00022	90.00	11306.	0.50	1.1935	1.1634	0.4170	0.3321	0.0849
70	0.00040	20.00	11306.	0.60	0.7031	0.7031	0.4310	0.3765	0.0695
71	0.00040	5.00	22500.	0.20	0.3331	0.2974	0.5830	0.3934	0.4896
72	0.00040	10.00	19000.	0.20	0.3691	0.6115	0.5200	0.4131	0.1149
73	0.00040	10.00	26000.	0.40	0.6357	0.6115	0.4054	0.1556	0.0274
74	0.00040	20.00	22500.	0.40	0.6723	0.6752	0.9390	0.9604	-0.0274
75	0.00040	20.00	26000.	0.60	0.9736	0.9226	0.7390	0.7886	-0.0496
76	0.00040	45.00	22500.	0.70	1.1124	1.1025	0.6390	0.7695	-0.1305
77	0.00040	90.00	19000.	0.60	0.9346	0.9419	0.5540	0.5636	-0.01316
78	0.00040	90.00	22500.	0.60	1.3493	1.4716	0.9630	1.6694	-0.7864
79	0.00040	90.00	22500.	0.90	1.2229	1.4716	0.6190	0.7792	-0.1532
80	0.00022	5.00	15000.	0.00	0.3012	0.2942	0.0550	0.3626	0.3252
81	0.00022	10.00	19000.	0.60	1.0077	0.9495	0.9390	0.3795	0.3545
82	0.00022	10.00	22500.	0.30	1.4000	1.4032	0.0370	0.6607	0.1463
83	0.00022	90.00	11306.	0.70	1.1559	1.0134	0.0030	0.0167	0.0123
84	0.00040	10.00	15000.	0.60	0.9457	0.9133	0.5940	0.4576	0.1370
85	0.00040	5.00	15000.	0.10	0.1317	0.1543	1.5470	0.5180	0.1020
86	0.00040	5.00	19000.	0.10	0.2540	0.2133	0.470	0.175	0.0720
87	0.00040	5.00	26000.	0.20	0.4463	0.4070	0.670	0.301	0.1399
88	0.00040	10.00	15000.	0.10	0.2157	0.1977	0.4170	0.5542	1.5528
89	0.00040	10.00	19000.	0.10	0.3773	0.2993	12.9700	17.5005	-4.5225
90	0.00040	10.00	26000.	0.20	0.5001	0.4311	0.5500	0.6000	0.4033
91	0.00040	10.00	13060.	0.10	0.1007	0.1007	1.479	0.0011	0.0011
92	0.00040	10.00	19000.	0.20	0.187	0.4099	0.440	0.0722	-0.0522
93	0.00040	10.00	26000.	0.20	0.9380	0.9133	1.840	0.0240	-0.0240
94	0.00040	10.00	11306.	0.10	0.2175	0.0330	0.2900	0.0292	-0.0292
95	0.00040	45.00	13060.	1.40	0.6953	0.7356	1.110	0.0370	-0.0370
96	0.00040	45.00	26000.	0.90	1.5452	1.6939	1.0000	1.0160	-0.0160
97	0.00040	90.00	11306.	0.10	0.2232	0.3390	5.4415	19.0739	-10.2029
98	0.00040	90.00	19000.	0.30	0.5653	0.9419	1.1650	1.9262	-0.7612
99	0.00040	90.00	26000.	0.30	1.6104	2.1947	1.4400	7.4627	-6.0227
100	0.00022	5.00	11306.	0.10	0.3000	0.2335	19.8590	6.1007	15.3363
101	0.00022	5.00	19000.	0.20	0.474	0.4034	0.6300	2.0539	3.9591
102	0.00022	5.00	22500.	0.30	0.6101	0.5004	0.440	1.5525	1.4905
103	0.00022	10.00	11306.	0.20	0.3600	0.3399	1.6590	1.0109	0.6731
104	0.00022	10.00	19000.	0.30	1.0201	0.9405	3.1400	1.9194	1.2306
105	0.00022	10.00	22500.	0.30	1.5041	1.4032	1.9040	1.4804	0.4236
106	0.00022	10.00	26000.	0.50	2.1300	0.1993	6.5400	6.0362	0.5038
107	0.00022	20.00	11306.	0.30	0.5453	0.5393	1.5600	1.3445	0.2155
108	0.00022	20.00	19000.	0.90	1.8135	1.9170	2.3400	3.8335	-0.9938
109	0.00022	45.00	11306.	0.50	0.3777	0.9629	1.3400	1.4394	-0.1494
110	0.00022	45.00	15000.	0.90	1.7244	1.9315	2.1290	3.9687	-1.6397
111	0.00022	90.00	11306.	0.60	1.1000	1.1634	1.6600	2.2627	-0.5967
112	0.00022	90.00	15000.	0.90	1.5052	2.5682	3.5600	14.5878	-11.0798
113	0.00040	5.00	11306.	0.20	0.4137	0.3477	3.4970	1.1768	2.3262
114	0.00040	10.00	11306.	0.30	0.5901	0.5364	2.4750	1.4332	1.0448
115	0.00040	10.00	19000.	0.30	1.0287	1.5616	1.5100	1.2555	0.2545
116	0.00040	20.00	11306.	0.60	0.9171	0.8651	1.6400	1.3152	0.3248
117	0.00040	20.00	15000.	0.90	1.7669	1.7759	2.4670	2.5233	-0.0563
118	0.00040	45.00	11306.	0.90	1.5495	1.5497	1.1000	1.1010	-0.0010
119	0.00040	90.00	11306.	0.90	1.7965	2.0179	2.7100	4.9876	-2.2776
120	0.000000E+01	6.000000E+01	2.047226E-02						

Figure 3-9 permits a quick evaluation of the quality of the functionalization of I_L . The abscissa scales represent the computed lethality integral, I_{LC} , (on logarithmic scale), and its corresponding computed survival ratio f_c given by Equation 3-46. The ordinate scales represent the residuals ΔI_L given by Equation 3-47, in linear scale, and the corresponding ratio $\frac{f}{f_c}$ of true survival ratio to computed survival ratio given by Equation 3-48.

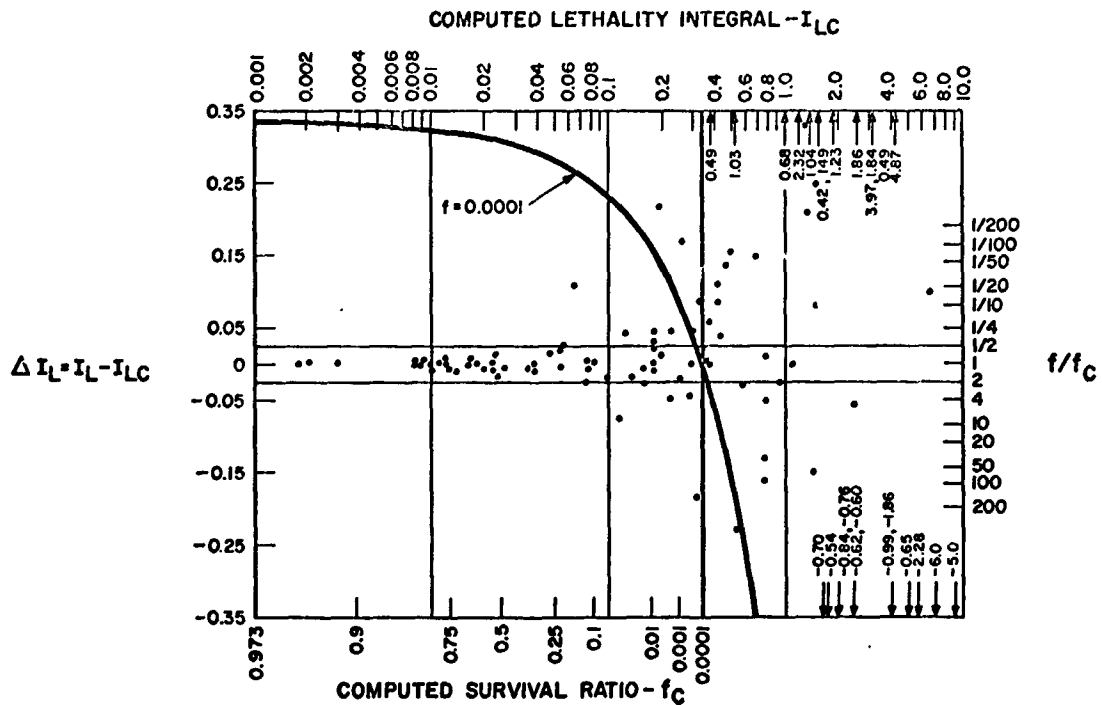


Figure 3-9. Evaluation of Functionalization

The locus of the points corresponding to f_c equal to 0.0001 is the vertical straight line. The locus of the points corresponding to f equal to 0.0001 is the curved line. For all the points which are simultaneously to the right of these two lines, the true kill as well as the computed kill are complete.

The two horizontal lines correspond to ratio $\frac{f}{f_c}$ equal to $\frac{1}{2}$ and 2. These lines define the bounds for all the points for which the true survival ratio f is within a factor 2 of the computed survival ratio f_c .

It can be readily seen from Figure 3-9 that most of the data points having survival ratio larger than 10^{-4} are within this band, that is, within this factor of 2 of the computed survival ratio. The few points which are not within the band would, however, almost all fall within a band having a factor of 4 from the computed survival ratio (f_c).

Since this was only a preliminary investigation, no special investigation was made of the few data points which were not within this band defined by a maximum factor of 4 between f_c and f .

3.14 CHANGES IN PARAMETERS

This preliminary investigation was conducted mainly to determine a procedure for functionalizing the lethality integral I_L . Since then, the following changes in the values or ranges of some parameters have appeared desirable:

- a. The range of ballistic coefficient, β , should cover from 4×10^{-6} to 4×10^{-3} slug/ft².
- b. The entry altitude, h_0 , should be much higher than the 721,000 feet we have used. A value of 2,000,000 feet seems to be satisfactory for the range of ballistic coefficients considered in item a, above.
- c. The range of entry angles, γ_E , should be extended. Viability of particles entering Mars atmosphere at a very small angle is of concern for the quarantine study.
- d. The solar absorptivity, α , should be varied to simulate nighttime as well as daytime entry into Mars atmosphere and to take into account experimental values of α which are much smaller than 1.
- e. While an initial (equilibrium) temperature of 500°R is reasonable for daytime entry, a lower initial temperature (183°R, for instance) should be used for nighttime entry.

The influence of these parameters and of these changes in parameter ranges is being investigated.

3.15 ORGANISMS CARRIED BY NONVIABLE PARTICLES

If we assume that, during all the entry trajectory, the microorganism attains the same temperature as the nonviable particle which carries it, the integral (I_L) can be computed for the particulate carrier (with its own solar absorptivity and its own emissivity) and inferences drawn or implied as to its effect on the microorganism.

Results are summarized in Table 3-4.

Table 3-4. I_L for Nonviable Particles in Full Sun

Material	$\downarrow V_E$	$\gamma_E = 5^\circ$		$\gamma_E = 45^\circ$		$\gamma_E = 90^\circ$	
		4×10^{-5}	4×10^{-4}	4×10^{-5}	4×10^{-4}	4×10^{-5}	4×10^{-4}
Aluminum	11 306	0.0006	14.11	0.0111	29.80	0.0438	26.28
	15 000	0.0012		2.034		2.856	
Fuzed Silica	11 306	0.0000	36.94	0.0315	42.16	0.1824	47.67
	15 000	0.0014		4.409		5.448	
Haynes-25	11 306	0.0000	9.49	0.0986	27.14	0.5332	26.9
	15 000	0.0010		3.128		4.047	
Magnesium	11 306	0.0001	28.54	0.0147	36.82	0.0518	38.89
	15 000	0.0022		2.791		3.651	
Epoxy Glass	11 306	5.46	125.20	5.71	68.41	4.66	75.86
	15 000						

For comparing Table 3-4 and Table 3-2, we can consider an equivalent emissivity, ϵ_Q , giving in Table 3-2 the same I_L as in Table 3-4 for the same combination (β , γ_E , V_E).

It can be seen that an equivalent emissivity, ϵ_Q , (between 0.1 and 0.3) can be defined for each of the first four materials. The equivalent emissivity increases with the entry angle γ_E : the materials seem to be less sensitive to the change in entry angle than the microorganism.

The conclusion would then be that any organism carried by the epoxy glass would receive a heat treatment which would kill it.

3.16 REFERENCES

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3.17 ACKNOWLEDGEMENTS

It is indeed a pleasure to acknowledge the contributions of the following persons to this investigation: J. Parker, who, with D. Florence, provided background information on his previous work, and by his knowledge helped considerably in the selection of the parameters, and, with N. Beerger, provided all the necessary computer runs as well as data on nonviable particles; C. Bursey, who made all the trajectory runs needed and, with D. Korenstein, pointed out the aerodynamic considerations which had to be taken into account; and M. Koesterer, who established the curve of thermal resistance of dry spores and explained how the curve was established. My thanks go also to Miss P. McManus who computed the lethality integral on the Desk Side Computer before P. Friend added an ad hoc subroutine to his thermodynamics computer program.

SECTION 4
SENSITIVITY STUDIES

by

E. Berger
R. Wolfson

TABLE OF CONTENTS

Section	Page
4 SENSITIVITY STUDIES	4-1
4.1 Introduction	4-1
4.2 Loose Particles	4-2
4.3 Attitude Control Gas	4-11
4.4 Acknowledgements	4-13

LIST OF ILLUSTRATIONS

Figure	Page
4-1 Planetary Quarantine Task Simplified Work Method Diagram	4-1
4-2 Detailed Computational Flow Diagram	4-3
4-3 Combination of Contamination Sources	4-5
4-4 Cumulative Probability of Viable Organisms	4-5
4-5 Probability Density Function of Loose Particles	4-6
4-6 Fraction of Viable Organisms on Particles of Different Sizes	4-7
4-7 Ejection Velocities	4-8
4-8 Probability of Entry for Various Time Intervals	4-9
4-9 Cumulative Probability of Viable Organisms	4-10
4-10 Initial Number of Viable Organisms in Attitude Control Gas System	4-11
4-11 Attitude Control Gas Use Profile	4-12
4-12 Size Distribution	4-12
4-13 Velocity Distribution	4-12
4-14 Start Analysis	4-13
4-15 Analysis Continued	4-13
4-16 Analysis Continued	4-13
4-17 Analysis Continued	4-13

SECTION 4
SENSITIVITY STUDIES

4.1 INTRODUCTION

The sensitivity studies involve the exercising of the basic math model with inputs from the basic studies as specifically applied to the Voyager mission and hardware. The initial cases shown are primarily designed to illustrate the process for working through the analysis and showing how the sensitivity of the contamination probability varies with different input parameters. In many cases the input parameters used have been essentially educated guesses. The continuing work in this area will use better and better input data, but these early studies serve to illustrate the areas which are important and require more careful analysis as the study proceeds.

Figure 4-1 is a simplified work flow diagram and illustrates how the sensitivity studies are related to the Quarantine Task.

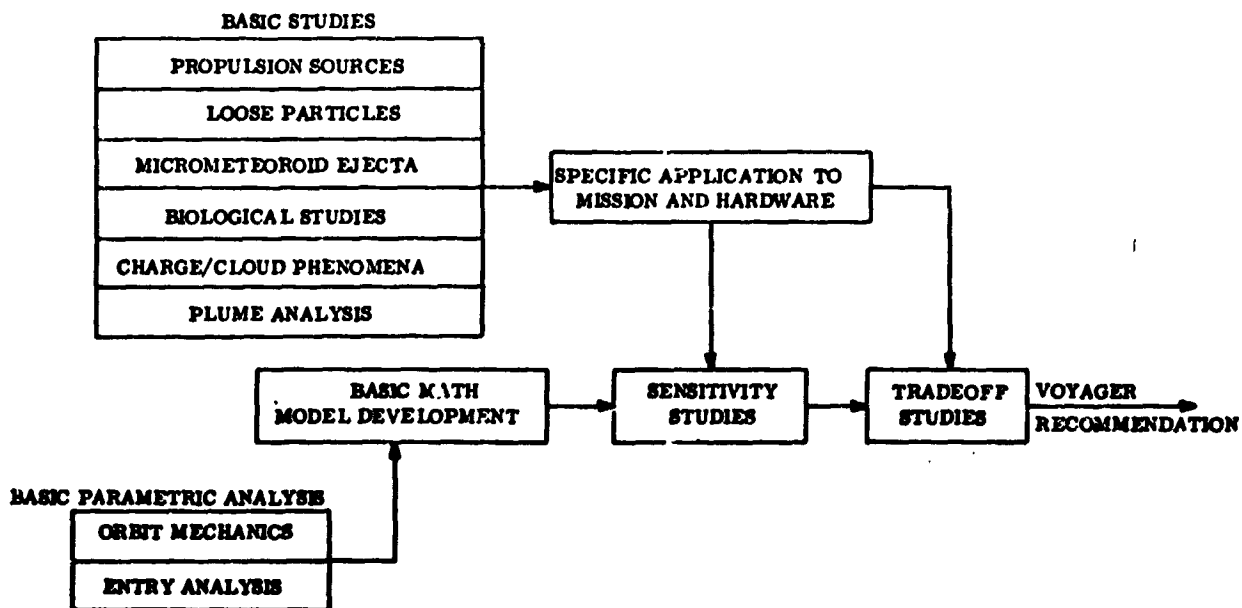


Figure 4-1. Planetary Quarantine Task Simplified Work Flow Diagram

The math model format, as illustrated in Figure 4-1, shows how the various sources of contamination are to be analyzed. The basic kill mechanisms are associated with columns. Each column either requires the development of input or operation on a particular portion of the math model.

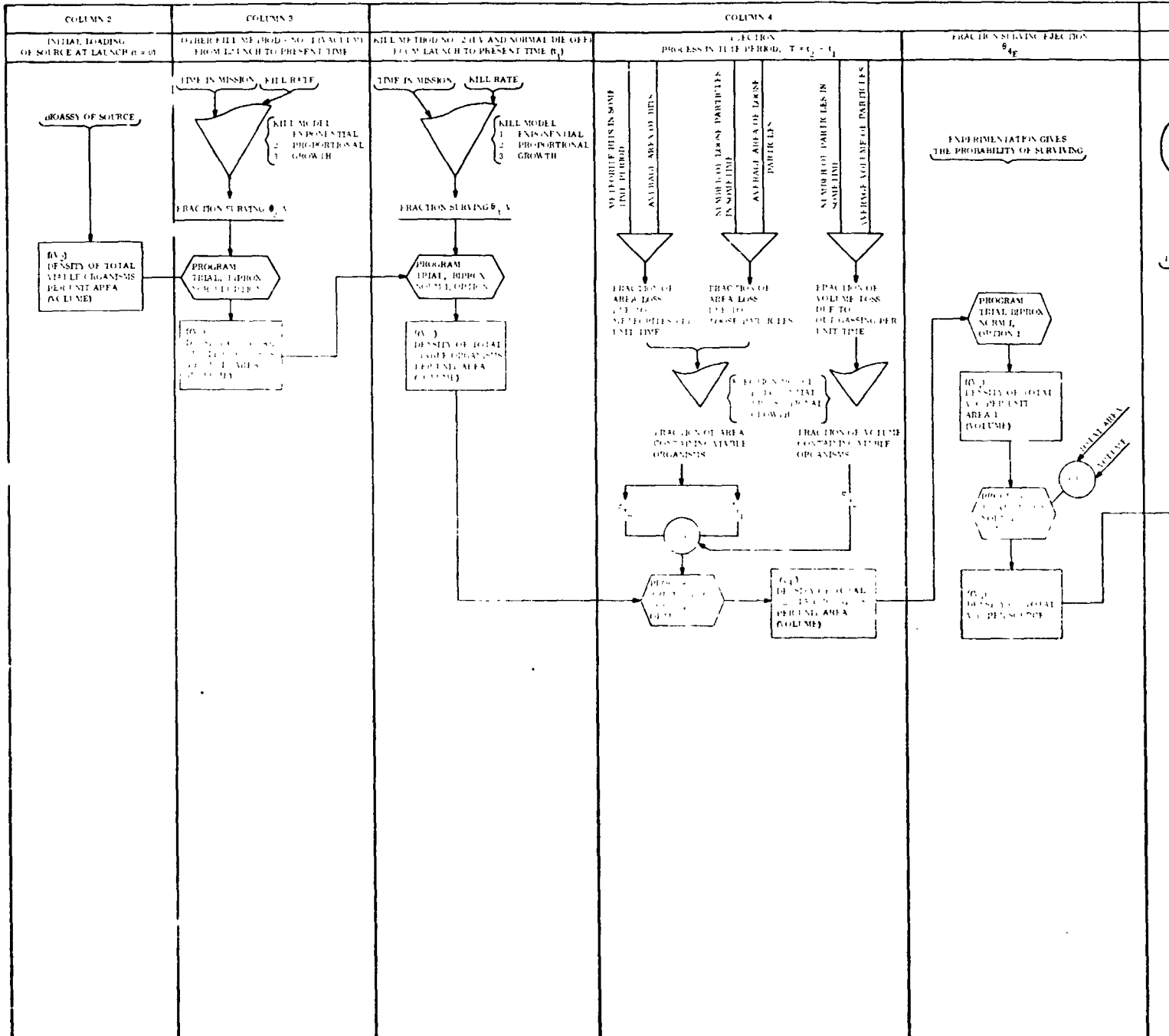
Figure 4-2 is a detailed computational flow diagram. This flow is for loose particles, micrometeoroid ejecta, and gaseous emissions, either cold or hot. In this figure rectangles give the output from or input to each column, and six-sided boxes tell what computer program is to be used.

Figure 4-3 illustrates how each of the contamination sources will be combined to give the probabilities of viable organisms reaching and growing on the planet.

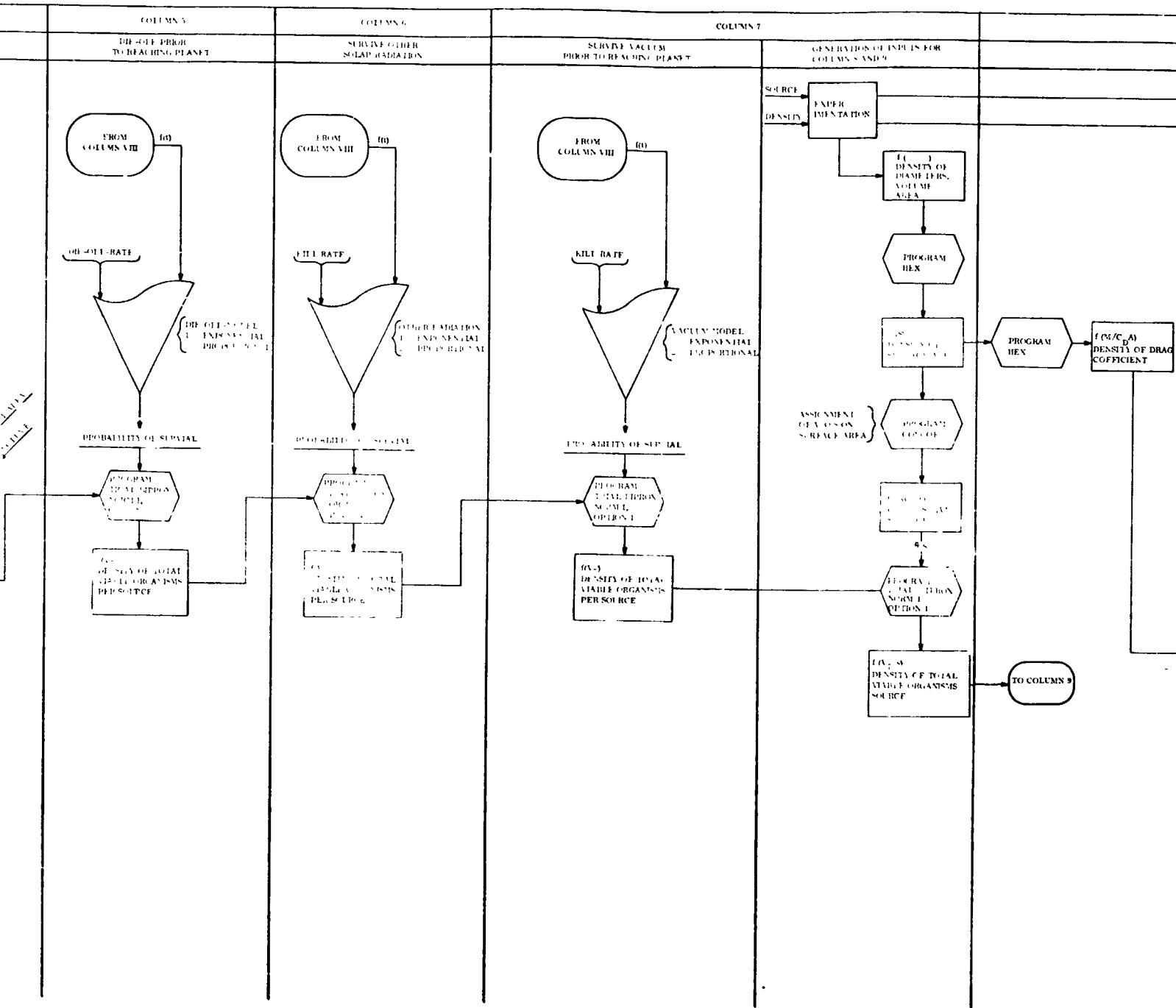
A few sources have been partially evaluated in a preliminary manner. None of the cases studied include all of the kill methods. The assumptions used will be stated with the results. Caution should be used in generalizing from the results presented. The assumptions must always be kept in mind.

4.2 LOOSE PARTICLES

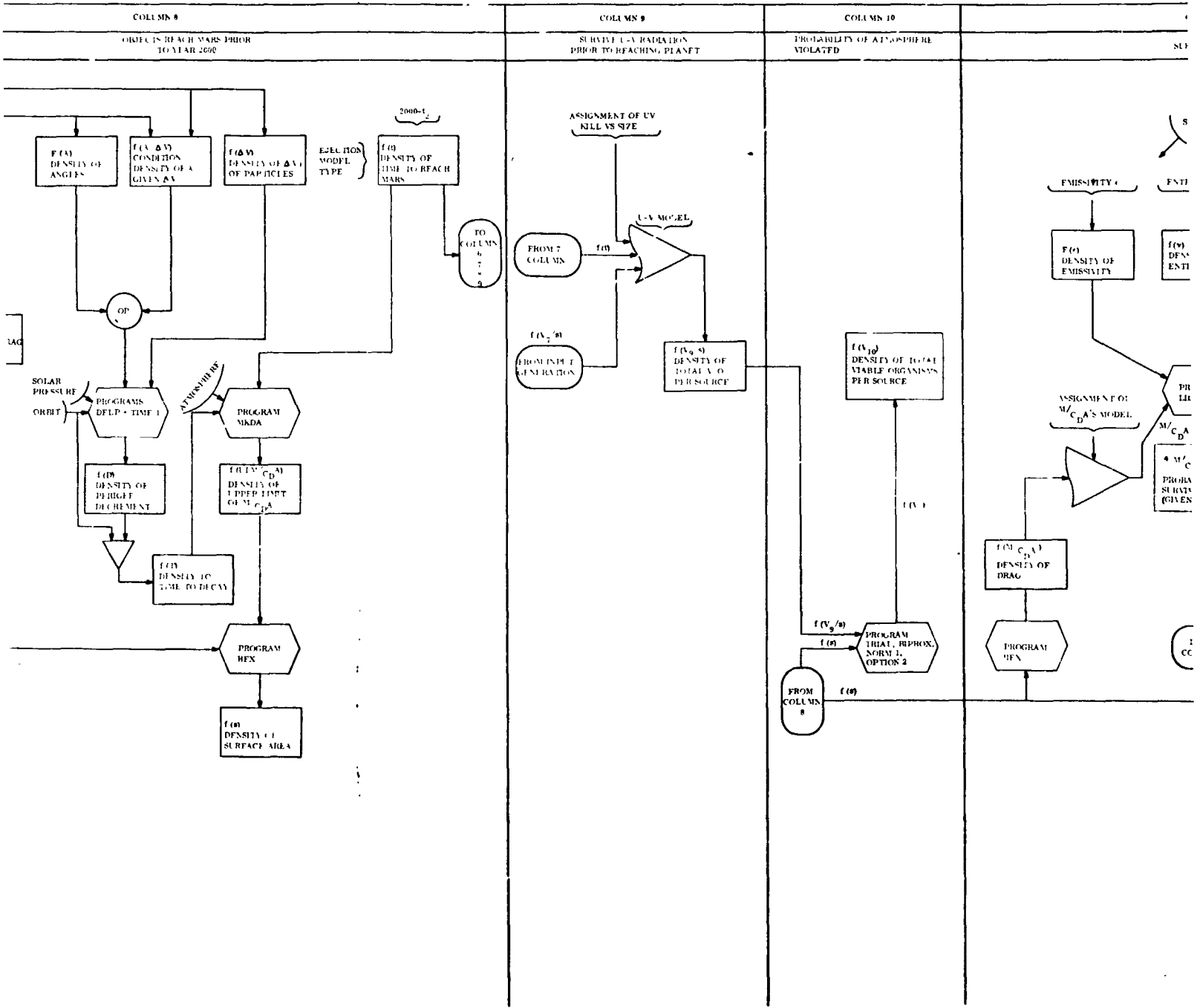
Preliminary results on loose particles are given in this section. The initial loading on the spacecraft was assumed as shown as curve (A) in Figure 4-4. Figure 4-4 shows the cumulative probability distribution function, whereas most of the other figures in this section are probability density functions. From basic data, VOY-C2-TM3, the total number of loose particles was estimated; then, with distribution of sizes, obtained by the same experimental investigation and modified to account for the lack of data below 150 microns, an estimate was made of the fraction of the total viable organisms on the particles and was found to be 0.001. This then gave the distribution labeled (B) in Figure 4-4 viable organisms on loose particles.



FOLD OUT FRAME 1



FOLD OUT FRAME # 2



Foldout FRAME 3

Figure 4-5 gives the distribution of particle diameters used in the loose particle investigation. An average weight per cubic foot of 68.6 was then assigned for the loose particles. Then, assuming a spherical relationship between diameter and surface area, the probability density

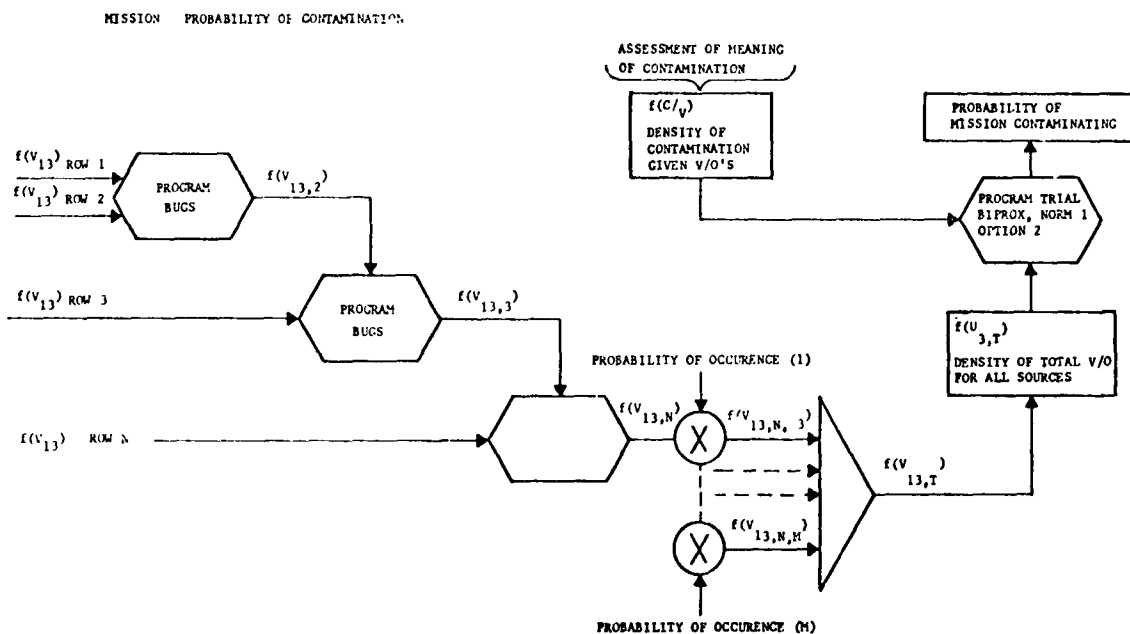


Figure 4-3. Combination of Contamination Sources

function of surface area was obtained by the use of program "HEX." Then, assuming that viable organisms were distributed in direct proportion to the surface area of the particle and that a loose particle below π square microns would not carry an organism (since particles below this size approach the size of a microorganism), we obtained the fraction of viable organisms on each range of surface area. This is an important step because the ballistic characteristic of each range of surface area is different and, to obtain a good estimate of the viable organisms surviving, these ballistic effects must be considered.

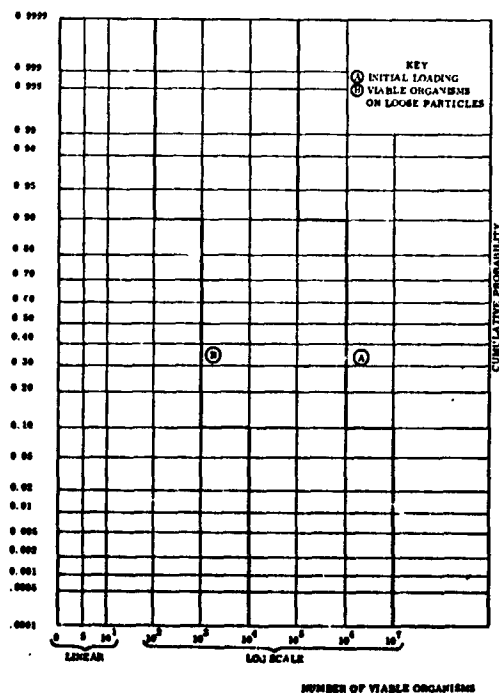


Figure 4-4. Cumulative Probability of Viable Organisms

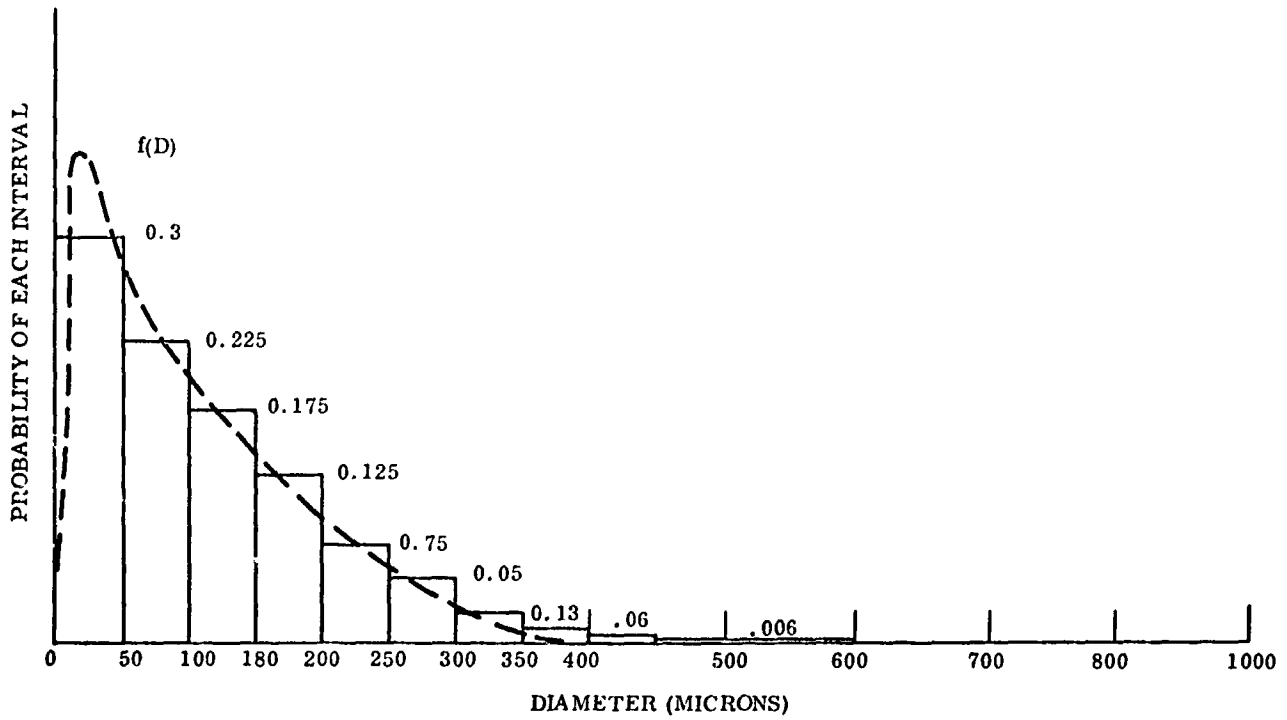


Figure 4-5. Probability Density Function of Loose Particles

Figure 4-6 gives the fraction of viable organisms on each surface area range considered. This assignment is accomplished with a program called "CONCOF." With the fraction of viable organisms assigned to each range of surface area and using the distribution of total viable organisms assigned to loose particles, a distribution of total viable organisms on each size is obtained.

The next step in the analysis of loose particles is to investigate both ejection prior to orbit insertion and ejection during the Mars orbiting phase. The analysis is only shown for those particles ejected during orbit. The number of loose particles carrying viable organisms has not been decreased by those leaving prior to reaching orbit insertion; this effect is subject to the micrometeoroid environment and will be investigated later. For now it is assumed that all loose particles come off in orbit and at the first apoapsis after insertion so that they will have a favorable time to decay from orbit to the planet.

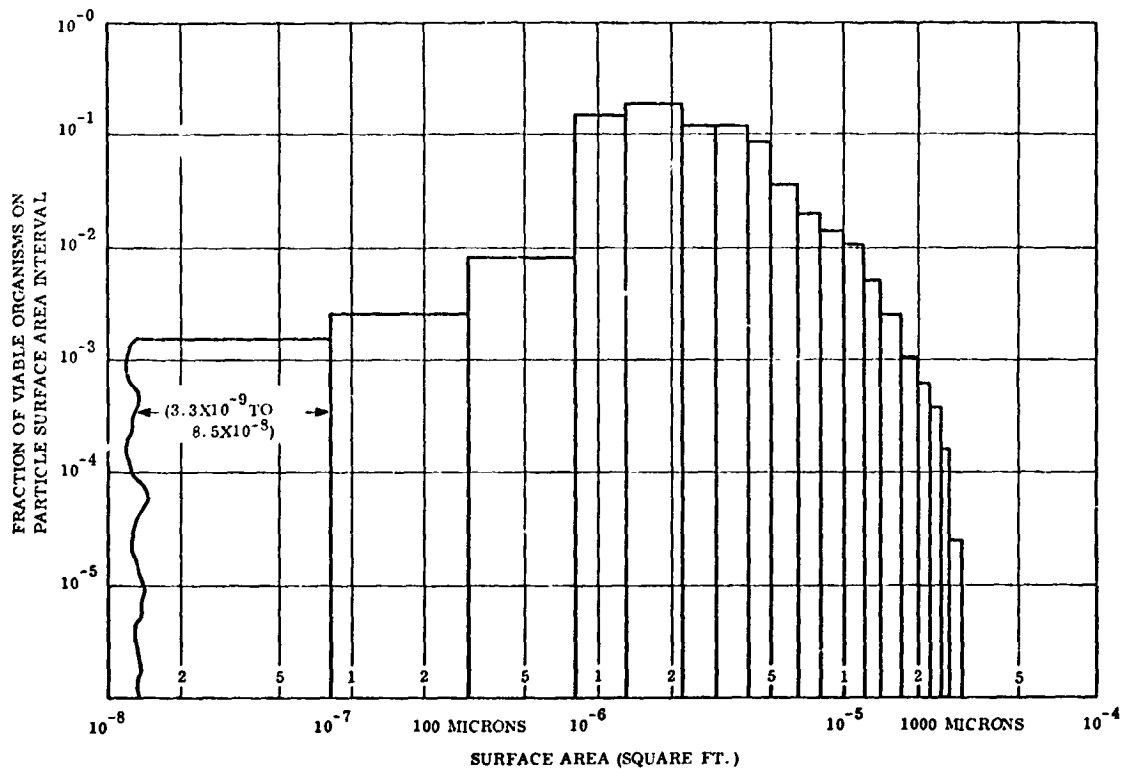


Figure 4-6. Fraction of Viable Organisms on Particles of Different Sizes

Under these assumptions, which are conservative, the analysis of orbit mechanics was performed for loose particles. Six orbits and two atmospheres were investigated.

Table 4-1 is a summary of the orbits and atmospheres investigated. All of the loose particles are assumed ejected at apoapsis which gives the largest decrease in periapsis altitude. The atmospheres are indicative of the expected variation in the VM-3 atmosphere as a function of solar heating of the planet. Atmosphere 1 is the night model; atmosphere 2 is the daylight model. Both models are less dense than the recent atmosphere adopted by JPL; consequently, more conservative from the quarantine viewpoint.

Table 4-1. Orbits and Atmospheres Investigated

Orbit	Atmosphere
1. 1000 x 10,000 km	VM-3 Atmos. Extended by Vachon
2. 500 x 10,000 km	1 for 0400 hours (min density)
3. 200 x 10,000 km	2 for 1400 hours (max density)
4. 1000 x 20,000 km	
5. 500 x 20,000 km	
6. 200 x 10,000 km	

Various ejection velocities and angles of leaving the spacecraft were assumed. The angles were assumed uniform over 4π steradians, and the velocity increment was that shown in Figure 4-7. These velocities are to be representative of loose particles drifting off of the spacecraft.

Programs DELP1, DELP2, and TIME1 were run to obtain the distribution of time to entry. Figure 4-8 gives the distribution for the six orbits and two atmospheres. Since atmosphere 1 is more conservative than the official JPL atmosphere, it is the one which was used in the investigation. Notice, however, that both orbit and atmosphere have significant effects. Notice also that only one-half of the loose particles have a chance of getting to the planet no matter what the orbit or atmosphere. This effect is due to the assumption of uniform angles of ejection; that is, one half of the particles leave at the wrong angles. This effect may be removed when solar pressure is included in the analysis. From Figure 4-8a it is seen that the probability of a particle reaching the planet prior to 30 years is 0.00121, for orbit type 1 and atmosphere 1.

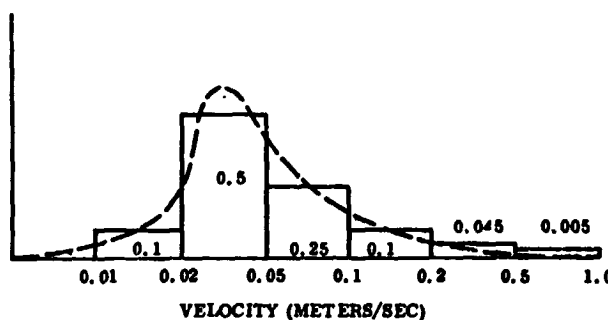


Figure 4-7. Ejection Velocities

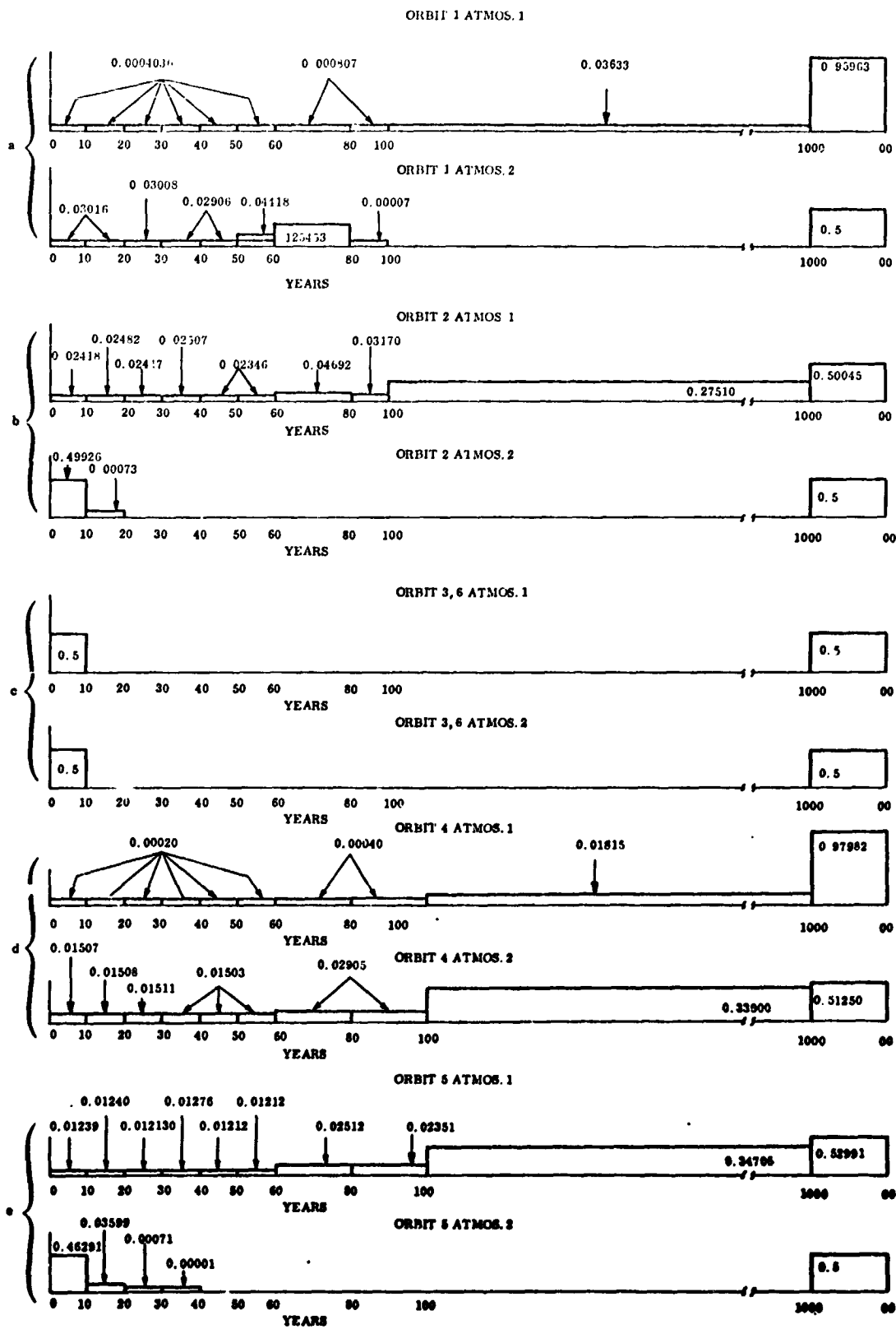


Figure 4-8. Probability of Entry for Various Time Intervals

Based on orbit 1 and atmosphere 1, programs M/C_DA and MCED were used to obtain the distribution of surviving sizes which enter within 30 years; this distribution was then used as an input to Program LID, which calculates the probability of surviving entry heating. The probability calculated was 0.1722.

The probabilities of surviving entry heating and entering prior to 30 years are then used on the distribution of total viable organisms on each size. The effect of U. V. kill and die-off are also assigned. The next result is then obtained by combining each size with its probability of occurrence. Figure 4-9, shown as a cumulative distribution function as was Figure 4-4, gives the preliminary estimate of this process using a probability of entering 0.00121, a probability of surviving heating of 0.1722, a probability of surviving U. V. of 0.1, and a probability of growth and spreading of 0.01.

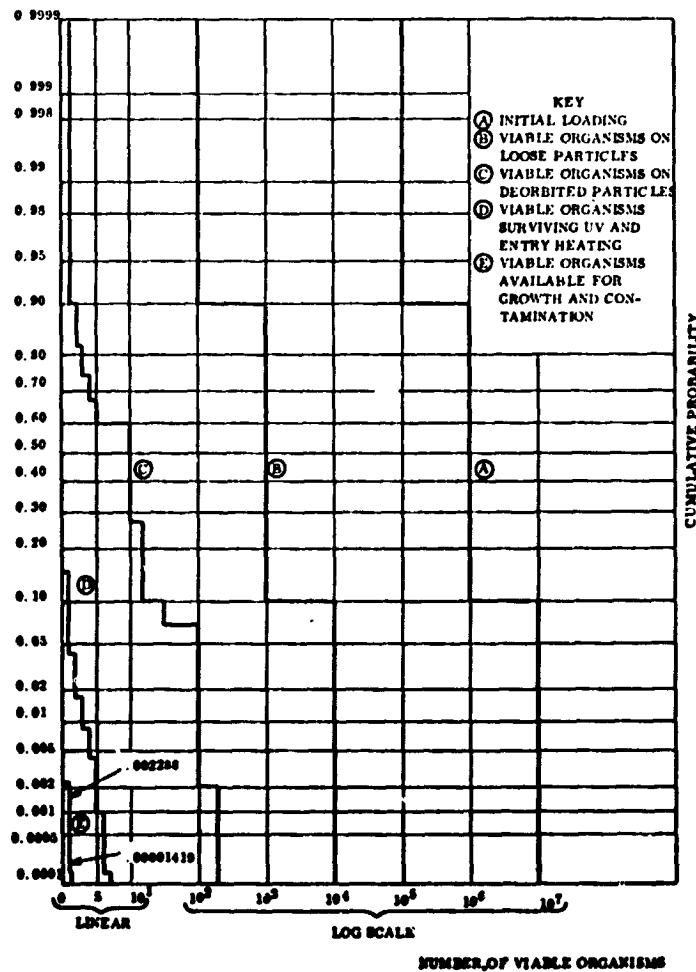


Figure 4-9. Cumulative Probability of Viable Organisms

When the final analysis is conducted, U. V. kill will be entered as a function of particle size. If the particle is 10 microns or smaller, kill will be assigned so that the viable organisms will have a high probability of being killed. The estimate of 0.1 is a "guesstimate" of the effects of U. V. The 0.01 growth and spreading probability is based on a recommendation presented to the last COSPAR meeting by Dr. C.W. Craven and J.O. Light of JPL.

The work in this section should be considered only as an example of the computational procedures and the way in which the quarantine problem can be studied. The actual data should not be used since the input data in many areas were guesstimates, and, in particular, the range of $M/C_D A$ values under consideration is currently being revised.

4.3 ATTITUDE CONTROL GAS

Figure 4-10 gives the initial number of viable organisms in the attitude control gas system. This initial number of viable organisms is assumed ejected in proportion to the usage rate of the attitude control gas as illustrated in Figure 4-11.

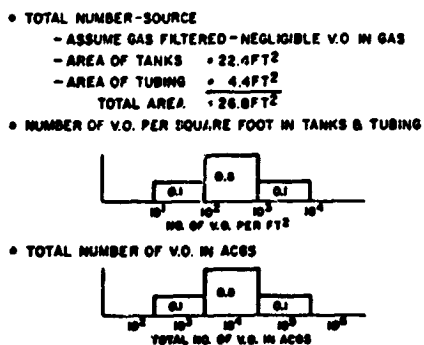


Figure 4-10. Initial Number of Viable Organisms in Attitude Control Gas System

Figures 4-12 and 4-13 give the parameters associated with the size distribution, drag parameter ($M/C_D A$), velocity, and ejection angles.

An analysis was conducted for a period of 1 day, 5 days before heliocentric encounter. An aim point based on the GE Task B study and a type I trajectory was assumed. The resulting probability of being on an impact trajectory was obtained as 0.00153. Figures 4-14 and 4-15 illustrate the results of applying first the fraction ejected and then the probability of being on an impact trajectory.

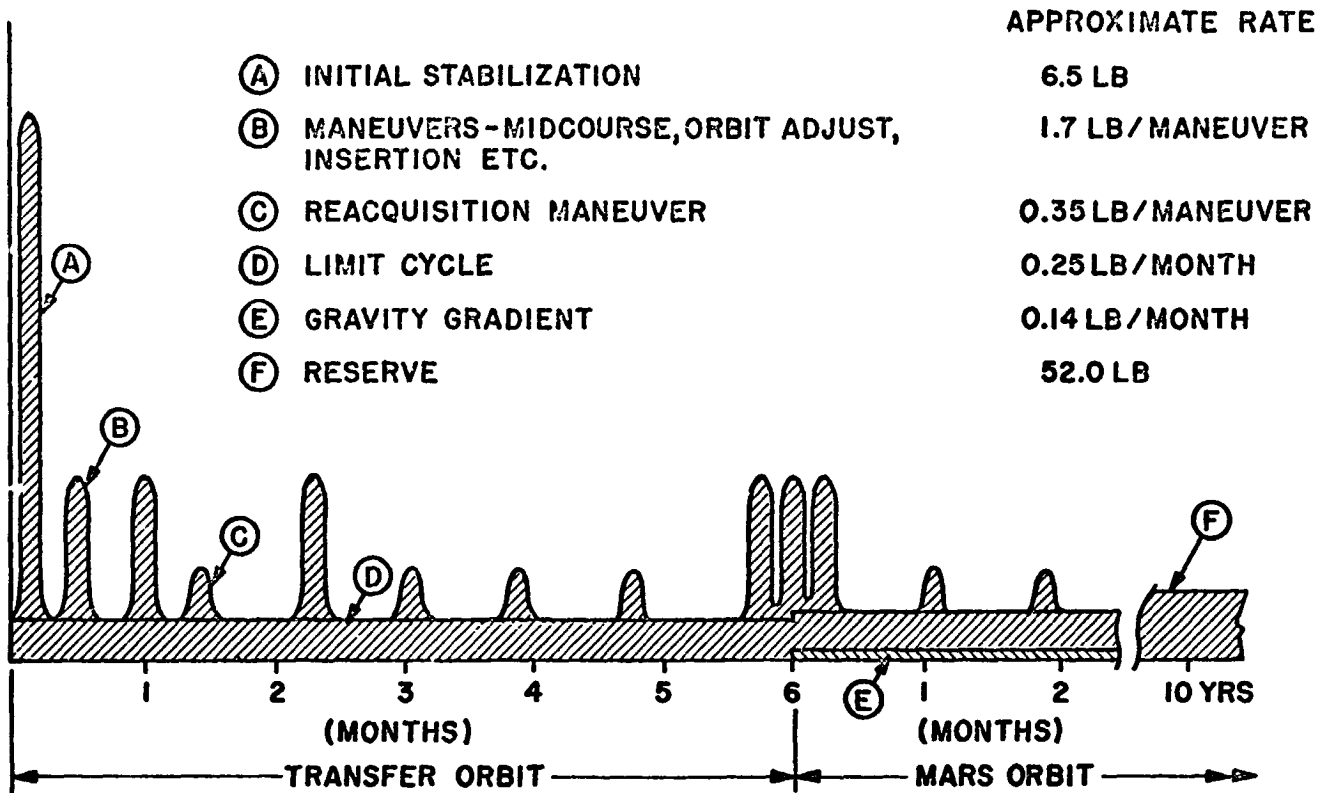


Figure 4-11. Attitude Control Gas Use Profile

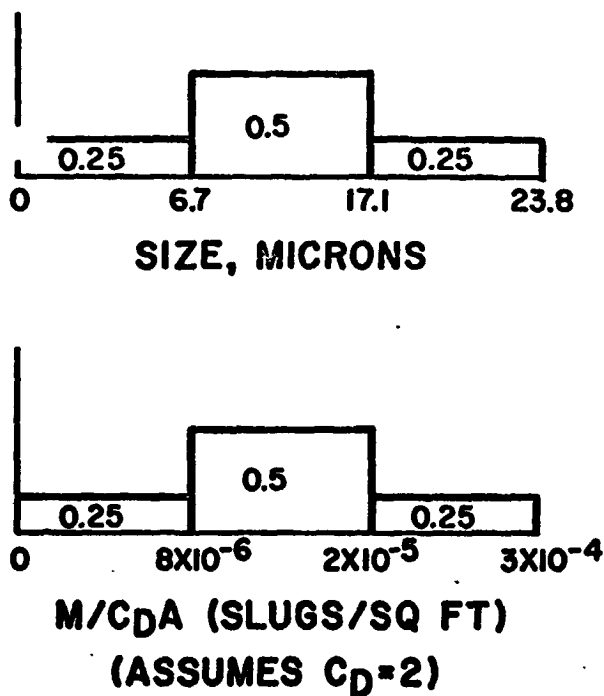


Figure 4-12. Size Distribution

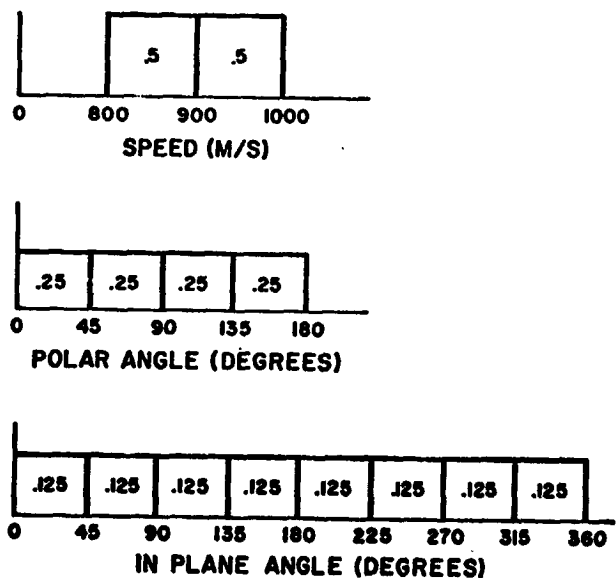


Figure 4-13. Velocity Distribution

124

ONE DAY - 5th DAY BEFORE ENCOUNTER
HELIOCENTRIC CASE

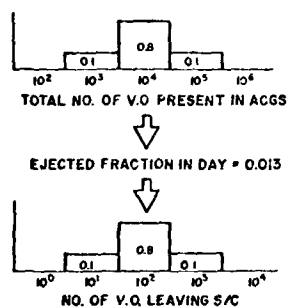
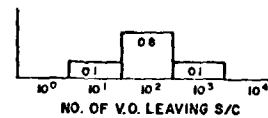


Figure 4-14. Start Analysis



PROBABILITY OF BEING ON IMPACT TRAJECTORY
= .00153

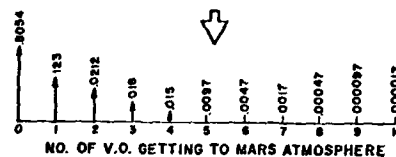


Figure 4-15. Analysis Continued

Figure 4-16 and 4-17 illustrate the results of the viable organisms surviving entry, U. V. kill and die-off (assumed to be 0.1), and growth and contamination (assumed to be 0.01).

Newly revised estimates of the $M/C_D A$ range required use of the entry heating program outside its original design range for this study. The range of accuracy of this program is currently being extended.

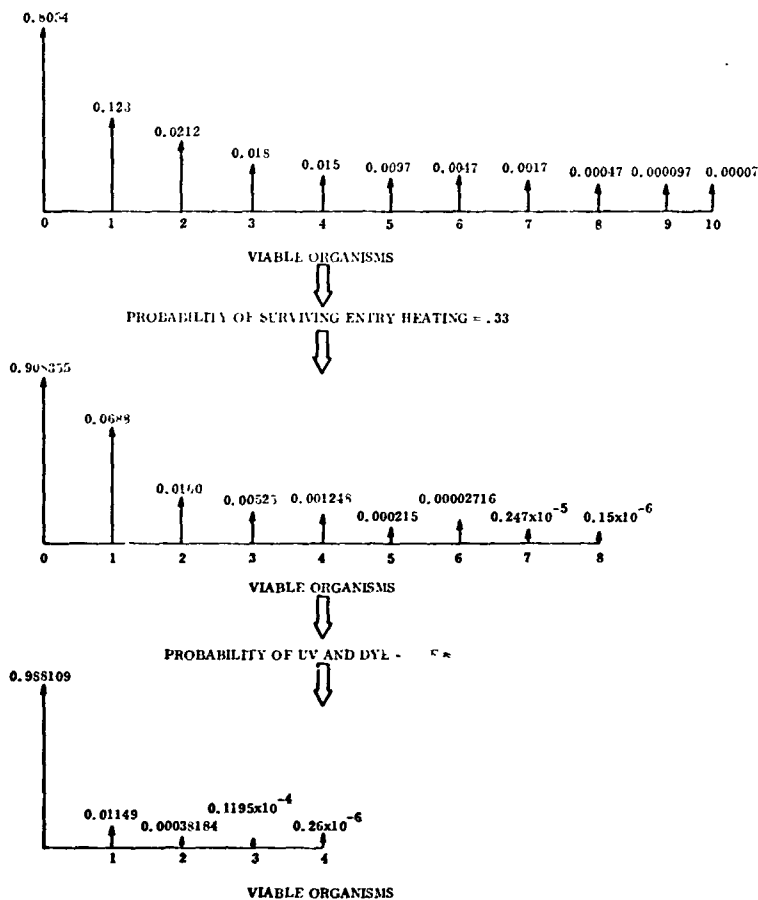


Figure 4-16. Analysis Continued

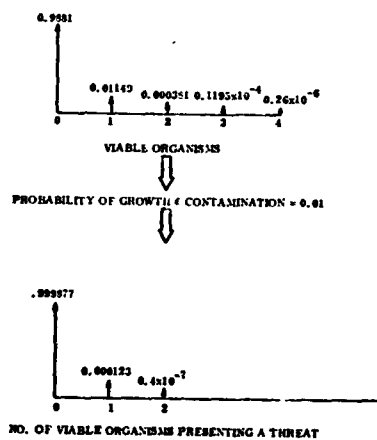


Figure 4-17. Analysis Continued

4.4 ACKNOWLEDGEMENTS

This is to acknowledge the assistance of Mr. D. Higley in operating the Desk Side Computer System and in reducing the computer output data for these sensitivity studies.

APPENDIX A
PROGRAM SOURCE LISTING

WAIT.

READY.

BASIC 1
1/16/67.

<LIST

```
00000 C *****BASIC COMBINATORIAL MODEL FOR VOYAGER*****
00010 COMMON GIN(100),PRO(100),G(100),GP(100),POUT(100),ETP(100)
00020 1000 FORMAT("1 BASIC COMBINATORIAL MODEL")
00030 1020 FORMAT("0 NUMBER, ROW END POINTS")
00040 1040 FORMAT("0 ROW PROBABILITIES")
00050 1060 FORMAT("0 NUMBER, STANDARD END POINTS")
00060 1080 FORMAT("1 READ IN CODE(1-5)")
00070 2000 FORMAT("0 FIRST LAST INPUT")
00080 + " OUTPUT")
00090 2020 FORMAT(" SIMPLE PROBABILITY")
00100 2040 FORMAT(" SCALE FACTOR")
00110 2060 FORMAT(" NUMBER, END POINTS FOR (",1PF14.5,")")
00120 2080 FORMAT(" CONDITIONAL PROBABILITIES")
00130 3000 FORMAT(" PROPORTION, PROBABILITY FOR (",1PE14.5,")")
00140
00150 PRINT 1000
00160 1 PRINT 1020
00170 READ:N1,(GIN(I),I=1,N1)
00180 N1M1=N1-1
00190 PRINT 1040
00200 READ:(PRO(I),I=1,N1M1)
00210 PRINT 1060
00220 READ:NST,(G(I),I=1,NST)
00230 NTM1=NST-1
00240 DO 20 I=1,NTM1
00250 GP(I)=0.
00260 DO 30 I=1,N1M1
00270 SP=GIN(I+1)-GIN(I)
00280 CALL LOAD(SP,PRO(I),GIN(I),GIN(I+1),NST,G,GP)
00290 TEST=100.
00300 DO 70 I=1,NTM1
00310 POUT(I)=0.
00320 PRINT 1080
00330 READ:ICODF
00340 GO TO (100,200,300,400,1),ICODF
00350 PRINT 2020
00360 READ:TR
00370 DO 150 I=2,NST
00380 IF(TEST-G(I))145,140,140
00390 NS=G(I)
00400 CALL BIACCUN(NS,TR,G,GP(I-1),POUT,NST)
00410 GO TO 150
00420 NG=I
00430 CALL RCONT(TR,G,POUT,NG,GP(I-1))
00440 CONTINUE
00450 GO TO 600
00460 DO 250 I=2,NST
00470 DO 210 J=1,NTM1
00480 GTP(I,J)=
```

PRINT ERROR

```

CALL RCOUNT(TR,G,POUT,NG,GP(I-1))
150 CONTINUE
   GO TO 600
200 DO 250 I=2,NST
   DO 210 J=1,NTM1
   GTP(J)=0.
210 PRINT 2060,G(I)
   READ:N2,(GIN(IZ),IZ=1,N2)
   N2M1=N2-1
   PRINT 2080
   READ:(PRO(IZ),IZ=1,N2M1)
   DO 220 K=1,N2M1
   SP=GIN(K+1)-GIN(K)
220 CALL LOAD(SP,PRO(K),GIN(K),GIN(K+1),NST,G,GTP)
   DO 230 L=1,NTM1
   POUT(L)=POUT(L)+GTP(L)*GP(I-1)
230 CONTINUE
   GO TO 600
300 DO 350 I=2,NST
   DO 310 J=1,NTM1
   CTP(J)=0.
310 PRINT 3000,G(I)
   READ:THET,PSI
   N2=3
   GIN(1)=G(I)
   GIN(2)=G(I)*THET
   GIN(3)=G(I)
   PRO(1)=PSI
   PRO(2)=1.-PSI
   DO 320 K=1,N2
   SP=GIN(K+1)-GIN(K)
320 CALL LOAD(SP,PRO(K),GIN(K),GIN(K+1),NST,G,GTP)
   DO 330 L=1,NTM1
   POUT(L)=POUT(L)+GTP(L)*GP(I-1)
330 CONTINUE
   GO TO 600
400 PRINT 2040
   READ:SCALE
   DO 410 I=1,NST
   G(I)=SCALE*G(I)
410 CONTINUE
   DO 500 I=1,NTM1
   POUT(I)=GP(I)
500 GP(I)=0.
   GO TO 600
600 PRINT 2000
   DO 620 I=1,NTM1
   PRINT:G(I),G(I+1),GP(I),POUT(I)
620 GP(I)=POUT(I)
   GO TO 600
END

```

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EXHIBIT FRAME

2

SLIST

```

00000 C *****SEGMENT BASIC*****
00010 SUBROUTINE RIACCI(KNS,TR,G,PROR,G0,NST)
00020 DIMENSION G(1),G0(1)
00030 FN=NS
00040 X=0.
00050 RAT=TR/(1.-TR)
00060 PR=(1.-TR)**NS
00070 GO TO 18
00080 5 IF(X-FN)10.,30.,30.
00090 10 PR=PR*(FN-X)*RAT/(X+1.)
00100 X=X+1.
00110 18 DO 25 I=2,NST
00120 IF(G(I)-X)25,19,19
00130 19 IF(G(I-1)-X)20,20,25
00140 20 G(I-1)=G(I-1)+PR*PROR
00150 GO TO 5
00160 25 CONTINUE.
00170 GO TO 5
00180 30 RETURN
00190 END
00200 SUBROUTINE LOAD(SP,TPR,ZL,ZH,M,Z,RP)
00210 DIMENSION Z(1),RP(1)
00220 IF(SP)100.,80.,20.
00230 DO 30 JL=2,M
00240 IF(Z(JL)-ZL)30.,30.,25
00250 25 ZLOW=ZL
00260 JTEMP=JL
00270 GO TO 40
00280 30 CONTINUE
00290 40 CONTINUE
00300 DO 60 J=JTEMP,M
00310 IF(Z(J)-ZH)50.,45.,45
00320 45 JSAVE=J
00330 ZHIGH=ZH
00340 GO TO 70
00350 50 ZHIGH=Z(J)
00360 RP(J-1)=RP(J-1)+((7HIGH-ZLOW)/SP)*TPR
00370 ZLOW=Z(J)
00380 60 CONTINUE
00390 70 RP(JSAVE-1)=RP(JSAVE-1)+((ZHIG-H-ZLOW)/SP)*TPR
00400 GO TO 100
00410 80 DO 90 J=2,M
00420 IF(ZL-Z(J-1))85.,82.,85
00430 82 RP(J-1)=RP(J-1)+TPR
00440 JSAVE=J
00450 GO TO 100
00460 85 IF(ZL-Z(J))82.,90.,90
00470 90 CONTINUE
00480 RP(M-1)=RP(M-1)+TPR
00490 JSAVE=M
00500 100 CONTINUE
00510 RETURN
00520 END
00530 SUBROUTINE RCONT(TR,G,G0,NG,PROR)
00540 DIMENSION G(1),G0(1),ANS(100)
00550 XMU=TR*(G(NG)
00560 SIGMA=SORTF(XMU*(1.-TR))
00570 IF(NG-2)50.,50.,130
00580 G1=(G(1)-XMU+.5)/SIGMA
00590 G2=(G(2)-XMU+.5)/SIGMA
00600 ANS(1)=HAD(G1,G2)

```

WOLDBOUT FR

```

SUBROUTINE RCONT(TR,G,GR,NG,PROR)
DIMENSION G(1),GR(1),ANS(100)
XMU=TR+G(NG)
SIGMA=SQRT(XMU*(1.-TR))
IF(NG-2)50,50,130
50 G1=(G(1)-XMU+.5)/SIGMA
G2=(G(2)-XMU+.5)/SIGMA
ANS(1)=HAD(G1,G2)
GO TO 190
130 DO 180 I=2,NG
IF(I-2)150,150,155
150 G1=(G(I-1)-XMU+.5)/SIGMA
G2=(G(I)-XMU)/SIGMA
GO TO 170
155 IF(I-NG)165,160,160
160 G1=(G(I-1)-XMU)/SIGMA
G2=(G(I)-XMU+.5)/SIGMA
GO TO 170
165 G1=(G(I-1)-XMU)/SIGMA
G2=(G(I)-XMU)/SIGMA
170 ANS(I-1)=HAD(G1,G2)
180 CONTINUE
190 TMP=0.
NRM1=NG-1
DO 200 I=1,NRM1
200 TMP=TMP+ANS(I)
DO 220 I=1,NRM1
220 G0(I)=G0(I)+PROR*ANS(I)/TMP
RETURN
END
FUNCTION HAD(X1,X2)
IF(ARCF(X1)-13.)20,20,10
10 X1 = SIGNF(13.0,X1)
20 IF(ARCF(X2)-13.)40,40,30
30 X2 = SIGNF(13.0,X2)
40 TEMP = 1.-41.42136
U1 = X1/TEMP
U2 = X2/TEMP
IF(U1)60,70,50
50 IF(U2)70,70,80
60 IF(U2)80,70,70
70 HAD = .5*(ERR(ARCF(U2))*SIGNF(1.,U2)-FRR(ARCF(U1))
1 *SIGNF(1.,U1))
RETURN
80 HAD = .5*(ERR(-ARCF(U1))*SIGNF(1.,U1)-ERR(-ARCF(U2))*
1 *SIGNF(1.,U2))
RETURN
END
C234567
FUNCTION FRC(W)
DIMENSION A(25),R(30)
M=24
A(1)=16443152242714E-13
A(2)=-9049760497548E-13
A(3)= 0643570883797E-13
A(4)= 0196418177368E-13
A(5)=-0001244215694E-13
A(6)=-0009101941905E-13
A(7)=-0001796219835E-13
A(8)= 0000139836786E-13
A(9)= 164789417E-13
A(10)= 39009267E-13
A(11)=- 00093145E-13
A(12)=- 03747896E-13
A(13)=- 01298818E-13

```



```

01050 A(1)=16443152242714E-13
01060 A(2)=-904976049754E-13
01070 A(3)= 0643570883797E-13
01080 A(4)= 019641817736E-13
01090 A(5)=-0001244215694E-13
01100 A(6)=-0009101941905E-13
01110 A(7)=-0001796219835E-13
01120 A(8)= 0000139836786E-13
01130 A(9)= 164789417E-13
01140 A(10)= 39009267E-13
01150 A(11)=- 00893145E-13
01160 A(12)=- 03747896E-13
01170 A(13)=- 01298818E-13
01180 A(14)=- 00136773E-13
01190 A(15)= 00077107E-13
01200 A(16)= 00046810E-13
01210 A(17)= 00011844E-13
01220 A(18)=- 0005F-13
01230 A(19)=- 1384E-13
01240 A(20)=- 0652E-13
01250 A(21)= 0145E-13
01260 A(22)= 0010E-13
01270 A(23)= 0024E-13
01280 A(24)= 0011E-13
01290 A(25)= 0002E-13
01300 X=ARSF(W)
01310 IF(ARSF(X)-.01) 100,110,110
01320 100 XERR = 2.0/(3.0+1.77245385)*X*(3.0-X**2)
01330 GO TO 140
01340 110 Z = (X-1.0)/(X+1.0)
01350 DO 120 I=1,30
01360 R(I)=0.
01370 120 CONTINUE
01380 DO 130 I=1,M
01390 M1=(M+1)-I
01400 R(M1)=2.0*Z*B(M1+1)-R(M1+2)+A(M1+1)
01410 130 CONTINUE
01420 F=3(2)+Z*B(1)+.5*A(1)
01430 XERR=1.-(1./1.77245385)*(EXP(-(X**2)))*F
01440 IF(ABS(X)-.01) 140,150,150
01450 140 CERR = 1.0-XERR
01460 GO TO 160
01470 150 CERR = (1.0/1.77245385)*(EXP(-(X**2)))*F
01480 160 IF(W) 180,170,170
01490 170 ERR = XERR
01500 GO TO 200
01510 180 ERR = CERR
01520 200 RETURN
01530 END

```

FORNITE BRAND

3

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SLIST
00000 COMMON X(100),Z(100),P(100),RP(100)
00010 PRINT 1000
00020 PRINT 1004
00030 FORMAT("9 FUNCTION CODE")
00040 PRINT 1005
00050 FORMAT(" 1, DIAMETER"/" 2, SURFACE AREA")
00060 PRINT 1006
00070 FORMAT(" 3, DRAG PARAMETER"/" 4, VOLUME"/" 5, MASS")
00080 PRINT 1007
00090 FORMAT(" 6, CROSS-SECTIONAL AREA")
00100 PRINT 1008
00110 PRINT 1010
00120 FORMAT("1, NUMBER,END POINT VALUES")
00130 READ:N1,X(1),I=1,N1)
00140 N1=N1-1
00150 PRINT 1015
00160 FORMAT(" PROBABILITIES")
00170 READ:(P(1),I=1,N1)
00180 PRINT 1005
00190 FORMAT(" GIVEN FUNCTION CODE, DENSITY, DRAG")
00200 READ:I1,D,C
00210 PRINT 1030
00220 PRINT 1030
00230 READ:I2
00240 IF(I2-7)4,1,1
00250 PRINT 1040
00260 FORMAT(" NUMBER, POINTS FOR NEXT DENSITY")
00270 READ:M,Z(I),I=1,M)
00280 CALL ADDER(N1,M,D,C,I1,I2)
00290 MT=M
00300 IF(MT-1)
00310 IF(RP(MT))6,6,7
00320 IF(MT-1)7,7,5
00330 JM=MT
00340 T=0.
00350 DO 10 I=1,JM
00360 T=RP(I)
00370 P(I)=RP(I)
00380 PRINT 1060
00390 FORMAT("RESULTING PROBABILITIES")
00400 PRINT 1070,I
00410 PRINT 1070,"CHECK SUM =",F10.6)
00420 N1=N1-JM
00430 DO 20 I=1,M
00440 X(I)=Z(I)
00450 PRINT 1001
00460 PRINT 1001
00470 .1=I2
00480 GO TO 2
00490 END
00500 SUBROUTINE ADDER(N1,M,D,C,I1,I2)
00510 COMMON X(100),Z(100),P(100),RP(100)
00520 DO 10 I=2,101
00530 RP(I-1)=0.
00540 DO 120 I=1,N1
00550 F1=HORN(I1,I2,X(I),D,C)
00560 F2=HORN(I1,I2,X(I+1),D,C)
00570 ZH=MAX(F1,F2)
00580 ZL=MIN(F1,F2)
00590 SP=ZH-ZL
00600 TPR=P(I)
00610 IF(SP)15,00,20
00620 PRINT 1

```

```

00540      J      1  KP(I-1)=0.
00550      DO 120 I=1,NIM1
00560      F1=HORN(I1,12,X(I),D,C)
00570      F2=HORN(I1,12,X(I+1),D,C)
00580      ZH=MAX(F1,F2)
00590      ZL=MIN(F1,F2)
00600      SP=ZH-ZL
00610      TPR=P(I)
00620      IF(SP)15,80,20
00630      15 PRINT 1500
00640      1500 FORMAT("INTERVAL NEGATIVE")
00650      GO TO 200
00660      20 DO 30 JL=2,M
00670      IF(Z(JL)-ZL)30,30,25
00680      ZLOW=ZL
00690      JTEMP=JL
00700      GO TO 40
00710      30 CONTINUE
00720      40 CONTINUE
00730      DO 60 J=JTEMP,M
00740      IF(Z(J)-ZH)50,45,45
00750      JSAVE=J
00760      ZHIGH=ZH
00770      GO TO 70
00780      50 ZHIGH=Z(J)
00790      RP(J-1)=RP(J-1)+((ZHIG-H-ZLOW)/SP)*TPR
00800      ZLOW=Z(J)
00810      60 CONTINUE
00820      70 RP(JSAVE-1)=RP(JSAVE-1)+((ZHIG-H-ZLOW)/SP)*TPR
00830      GO TO 120
00840      80 DO 90 J=2,M
00850      IF(ZL-Z(J-1))85,82,85
00860      82 RP(J-1)=RP(J-1)+TPR
00870      JSAVE=J
00880      GO TO 120
00890      85 IF(ZL-Z(J))82,90,90
00900      90 CONTINUE
00910      RP(M-1)=RP(M-1)+TPR
00920      JSAVE=M
00930      120 CONTINUE
00940      200 RETURN
00950      END
00960      FUNCTION HORN(INUTS,NUITS, D,C)
00970      IX=INUTS
00980      JX=NUITS
00990      PI=3.14159265
01000      X=Y
01010      1 GO TO (10,20,30,40,50,60),IX
01020      10 X=PI*X
01030      11 IX=IX+1
01040      16 X=1
01050      18 IF(IX-JX)1,80,1
01060      20 X=(2.*D*SORTF(X/PI))/(3.*C)
01070      GO TO 11
01080      30 X=(9.*PI/16.)*(C*X/D)**3
01090      GO TO 11
01100      40 X=X*D
01110      GO TO 11
01120      50 X=(PI/4.)*(6.*X/(PI*D))**(2./3.)
01130      GO TO 11
01140      60 X=SORTF(4.*X/PI)
01150      GO TO 11
01160      80 HORN=X
01170      RETURN
01180      END

```

TRACE 2

```
00000 COMMON X(100),Y(100),Z(100),P(100),FP(100),RP(100)
00010 PRINT 1000
00020 FORMAT("PROGRAM TO FIND PROBABILITY OF SUMS")
00030 PRINT 1010
00040 FORMAT("1 NUMBER,POINTS FOR FIRST DENSITY")
00050 READ:N1,(X(I),I=1,N1)
00060 N1M1=N1-1
00070 PRINT 1015
00080 FORMAT("FIRST SET OF PROBABILITIES")
00090 READ:(P(I),I=1,N1M1)
00100 PRINT 1020
00110 FORMAT("NUMBER,POINTS FOR RESULTING DENSITY")
00120 READ:M,(Z(I),I=1,M)
00130 2 PRINT 1030
00140 FORMAT("NUMBER,POINTS FOR NEXT DENSITY")
00150 READ:N2,(Y(I),I=1,N2)
00160 PRINT 1040
00170 FORMAT("NEXT SET OF PROBABILITIES")
00180 N2M1=N2-1
00190 READ:(PP(I),I=1,N2M1)
00200 CALL ADDER(N1M1,N2M1,M,JSAVE)
00210 JM1=JSAVE-1
00220 T=0.
00230 DO 10 I=1,JM1
00240 T=T+PP(I)
00250 10 P(I)=NP(I)
00260 PRINT 1060
00270 FORMAT("RESULTING PROBABILITIES")
00280 PRINT:(RP(I),I=1,JM1)
00290 PRINT 1070,T
00300 FORMAT("CHECK SUM =",F10.6)
00310 N1M1=JM1
00320 DO 20 I=1,M
00330 20 X(I)=Z(I)
00340 PRINT 1001
00350 1001 FORMAT(IH1)
00360 GO TO 2
00370 END
00380 SUBROUTINE ADDER(N1M1,N2M1,M,JSAVE)
00390 COMMON X(100),Y(100),Z(100),P(100),FP(100),RP(100)
00400 DO 10 I=2,M
00410 10 RP(I-1)=0.
00420 DO 120 K=1,N1M1
00430 DO 100 I=1,N2M1
```

OLD DOT FROM

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SUBROUTINE ADDEK(N1M1,N2M1,M,JSAVE)
COMMON X(100),Y(100),Z(100),F(100),FP(100),RP(100)
DO 10 I=2,M
  RP(I-1)=0.
DO 120 K=1,N1M1
DO 100 I=1,N2M1
  ZL=X(K)+Y(I)
  ZH=X(K+1)+Y(I+1)
  SP=ZH-ZL
  TPK=FP(K)*RP(I)
  IF(SP)15,80,20
15 PRINT 1500
1500 FORMAT("INTEKVAL NEGATIVE")
20 DO 30 JL=2,M
  IF(Z(JL)-ZL)30,30,25
25 ZLOW=ZL
  JTEMP=JL
  GO TO 40
30 CONTINUE
40 CONTINUE
DO 60 J=JTEMP,M
  IF(Z(J)-ZH)50,45,45
45 JSAVE=J
  ZHIGH=ZH
  GO TO 70
50 ZHIGH=Z(J)
  RP(J-1)=RP(J-1)+((ZHIGH-ZLOW)/SP)*TPK
  ZLOW=Z(J)
  CONTINUE
70 RP(JSAVE-1)=RP(JSAVE-1)+((ZHIGH-ZLOW)/SP)*TPK
  GO TO 100
80 DO 90 J=2,M
  IF(ZL-Z(J-1))85,82,85
82 RP(J-1)=RP(J-1)+TPK
  JSAVE=J
  GO TO 100
85 IF(ZL-Z(J))82,90,90
90 CONTINUE
  RP(M-1)=RP(M-1)+TPK
  JSAVE=M
100 CONTINUE
120 CONTINUE
200 RETURN
END

```

EMERGENCY PAGE 2

BINOM X
1/13/67

```

000000 C      BINOMX-----NORMAL APPROXIMATION TO NORMAL
000100 1 PRINT 2000
000200 20000 FORMAT("1  NORMAL APPROXIMATION TO BINOMIAL")
000300 2 PRINT 2000
000400 20000 FORMAT("0READ FIRST, LAST VALUE, NUMBER, PROBABILITY")
000500 READ: X1, X2, XMAX, THETA
000600 XMU=XMAX*THETA
000700 SIGMA=SQRTF(XMU*(1.-THETA))
000800 X1=(X1-.5-XMU)/SIGMA
000900 X2=(X2+.5-XMU)/SIGMA
001000 ANS=HAD(X1, X2)
001100 PRINT 2025, X1, X2
001200 2025 FORMAT(" STANDARD NORMAL LIMITS*****", 2E14.6)
001300 PRINT 2030, ANS
001400 2030 FORMAT(" PROBABILITY=", E14.6)
001500 GO TO 2
001600 END
001700 FUNCTION HAD(X1, X2)
001800 TEMP=1.4142136
001900 U1=X1/TEMP
002000 U2=X2/TEMP
002100 IF(U1) 200, 200, 100
002200 IF(U2) 200, 200, 900
002300 IF(U2) 900, 200, 200
002400 HAD=.5*(ERR(ARSF(U2))*SIGNF(1., U2)-ERR(ARSF(U1)))
002500 1 *SIGNF(1., U1))
002600 RETURN
002700 HAD=.5*(ERR(-ARSF(U1))*SIGNF(1., U1)-ERR(-ARSF(U2)))*
002800 1 SIGNF(1., U2))
002900 RETURN
003000 END
003100 FUNCTION ERR(W)
003200 DIMENSION A(25), B(30)
003300 M=24
003400 A(1)=1644315222714E-13
003500 A(2)=-9049760497548E-13
003600 A(3)= 0643570883797E-13
003700 A(4)= 0196418177368E-13
003800 A(5)=-0001244215694E-13
003900 A(6)=-0009101941905E-13
004000 A(7)=-0001796219835E-13
004100 A(8)= 0000139836786E-13
004200 A(9)= 1.7717E-13

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10718

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00370 A(4)= 0196418177368E-13
00380 A(5)=-000124215694E-13
00390 A(6)=-0009101941905E-13
00400 A(7)=-0001796219835E-13
00410 A(8)= 000139836786E-13
00420 A(9)= 164789417E-13
00430 A(10)= 39009267E-13
00440 A(11)=- 00893145E-13
00450 A(12)=- 03747896E-13
00460 A(13)=- 01298818E-13
00470 A(14)=- 00136773E-13
00480 A(15)= 00077107E-13
00490 A(16)= 00046810E-13
00500 A(17)= 00011844E-13
00510 A(18)=- 00005E-13
00520 A(19)=- 1384E-13
00530 A(20)=- 0652E-13
00540 A(21)= 0145E-13
00550 A(22)= 0010E-13
00560 A(23)= 0024E-13
00570 A(24)= 0011E-13
00580 A(25)= 0002E-13
00590 X=ARSF(W)
00600 IF(ARSF(X)-.01)1,2,2
00610 1 XERR=2.0/(3.0*1.77245385)*X*(3.0-X**2)
00620 2 GO TO 6
00630 3 Z=(X-1.0)/(X+1.0)
00640 4 DO 3 I=1,30
00650 5 R(I)=0.
00660 6 CONTINUE
00670 7 DO 4 I=1,M
00680 8 M1=(M+1)-I
00690 9 R(M1)=2.0*Z*B(M1+1)-R(M1+2)+A(M1+1)
00700 10 CONTINUE
00710 11 F=-R(2)+Z*R(1)+.5*A(1)
00720 12 XERR=1.0-(1.0/1.77245385)*(FXPF(-(X**2)))*F
00730 13 IF(ARSF(X)-.01)6,7,7
00740 14 CERR=1.0-XERR
00750 15 GO TO 5
00760 16 CERR=(1.0/1.77245385)*(EXP(-(X**2)))*F
00770 17 IF(W)9,8,8
00780 18 ERR=XERR
00790 19 GO TO 13
00800 20 ERR=CERR
00810 21 RETURN
00820 22 END

```

NOLOGUE TRACE

2

PLOAD
1/13/67

```
00000      COMMON X(100),Y(100),Z(100),RP(100),M
00010      PRINT 2000
00020 2000 FORMAT("1  PROGRAM TO LOAD PROBABILITIES")
00030      PRINT 2020
00040 2020 FORMAT("0NUMBER, OUTPUT GRID VALUES")
00050      1 READ:M,(Z(I),I=1,M)
00060      2 PRINT 2040
00070 2040 FORMAT(" INTERVAL START, END, PROBABILITY")
00080      READ:ZL,ZH,TPR
00090      CALL ADDER(ZL,ZH,TPR)
00100      PRINT 2060
00110 2060 FORMAT("0      RESULTING PROBABILITIES")
00115      MM1=M-1
00120      PRINT:(RP(I),I=1,MM1)
00130      PRINT 1000
00140 1000 FORMAT (1H1)
00150      GO TO 2
00160      END
00170      SUBROUTINE ADDER(ZL,ZH,TPR)
00180      COMMON X(100),Y(100),Z(100),RP(100),M
00190      DO 10 I=2,M
00200      10 RP(I-1)=0.
00210      SP=ZH-ZL
00220      IF(SP)15,80,20
00230      15 PRINT 1500
00240 1500 FORMAT("0 INTERVAL NEGATIVE")
00250      GO TO 200
00260      20 DO 30 JL=2,M
00270      IF(Z(JL)-ZL)30,30,25
00280      25 ZLOW=ZL
00290      JTEMP=JL
00300      GO TO 40
00310      30 CONTINUE
00320      40 CONTINUE
00330      DO 60 J=JTEMP,M
00340      IF(Z(J)-ZH)50,45,45
00350      45 JSAVE=J
00360      ZHIGH=ZH
00370      GO TO 70
00380      50 ZHIGH=Z(J)
00390      RP(J-1)=RP(J-1)+((ZHIGH-ZLOW)/SP)*TPR
00400      ZLOW=Z(J)
00410      60 CONTINUE
00420      70 RP(JSAVE-1)=RP(JSAVE-1)+((ZHIGH-ZLOW)/SP)*TPR
00430      GO TO 100
00440      80 DO 90 J=2,M
00450      IF(ZL-Z(J-1))85,82,85
00460      82 RP(J-1)=RP(J-1)+TPR
00470      GO TO 100
00480      85 IF(ZL-Z(J))82,90,90
00490      90 CONTINUE
00500      RP(M-1)=RP(M-1)+TPR
00510      100 CONTINUE
00520      120 CONTINUE
00530      200 RETURN
00540      END
```



```

00000 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
00010 COMMON A(20),B(20),C(20),NN,YMEAN
00020 PRINT 1000
00030 PRINT 999
00040 999 FORMAT("1 ORBIT TYPE(1-6),ATMOS TYPE(1 OR 2)")
00050 1000 FORMAT(" TIME IN MARS ORBIT PROGRAM")
00060 NN=4
00070 HEAD:11,111
00080 A(1)=A(2)=A(3)=A(4)=600.
00090 B(1)=6.
00100 B(2)=66666.67
00110 B(3)=51333.33
00120 B(4)=46285.71
00130 GO TO (60,80),111
00140 C(1)=.01961519
00150 C(2)=-4.065703E-5
00160 C(3)=8.695401E-8
00170 C(4)=-1.711093E-10
00180 YMEAN=-.1833621
00190 GO TO 90
00200 C(1)=.0157079
00210 C(2)=-2.496592E-5
00220 C(3)=5.334886E-8
00230 C(4)=-1.376223E-10
00240 YMEAN=-3.913773
00250 GO TO (91,92,93,94,95,96),11
00260 91 HA=10000.
00270 PA=1000.
00280 GO TO 100
00290 92 HA=10000.
00300 PA=500.
00310 GO TO 100
00320 93 HA=10000.
00330 PA=200.
00340 GO TO 100
00350 94 HA=20000.
00360 PA=1000.
00370 GO TO 100
00380 95 HA=20000.
00390 PA=500.
00400 GO TO 100
00410 96 HA=20000.
00420 PA=200.
00430 100 PRINT 998
00440 998 FORMAT(" APOAPSIS(KM) PERIAPSIS(KM)")
00450 PRINT:HA,PA
00520 2 PRINT 1010
00530 1010 FORMAT(" NUMBER, PERIAPSIS VALUES")
00540 READ:N,(X(I),I=1,N1)
00550 N1=N1-1
00560 PRINT 1015
00570 1015 FORMAT(" PERIAPSIS PROBABILITIES")
00580 READ:(P(I),I=1,N1M1)
00590 PRINT 1020
00600 1020 FORMAT(" NUMBER, VALUES FOR TIME IN ORBIT")
00610 READ:M,(Z(I),I=1,M)
00620 PRINT 1030
00630 1030 FORMAT(" NUMBER, VALUES FOR DRAG PARAMETER")
00640 READ:Y,(I=1,11)

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REVERSE ENGINEERING

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00570 FORMAT(" PERIAPSIS PROBABILITIES")
00580 READS(P(I),I=1,N1M1)
00590 PRINT 1020
00600 FORMAT(" NUMBER, VALUES FOR TIME IN ORBIT")
00610 READS(Z(I),I=1,M)
00620 PRINT 1030
00630 FORMAT(" NUMBERS, VALUES FOR DRAG PARAMETER")
00640 READ:N2,(Y(I),I=1,N2)
00650 PRINT 1040
00660 FORMAT(" DRAG PARAMETERS PROBABILITIES")
00670 N2=N2-1
00680 READ:(P(I),I=1,N2M1)
00690 CALL MULPY(N1M1,N2M1,M,JSAVE,HA)
00700 MT=M
00710 5 MT=MT-1
00720 IF(KF(MT))6,6,7
00730 6 IF(MT-1)7,7,5
00740 7 JM1=MT
00750 T=0.
00760 DO 10 I=1,JM1
00770 T=T+KP(I)
00780 PRINT 1060
00790 FORMAT(" TIME IN ORBIT DISTRIBUTION")
00800 PRINT 1070,T
00810 1070 FORMAT("CHECK SUM =",F10.6)
00820 PRINT 1001
00830 1001 FORMAT(1M1)
00840 GO TO 2
00850 END
00860 SUBROUTINE MULPY(N1M1,N2M1,M,JSAVE,HA)
00870 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
00880 COMMON A(20),B(20),C(20),NN,YMEAN
00890 DO 10 I=2,M
00900 10 KP(I-1)=0.
00910 DO 120 K=1,N1M1
00920 DO 100 I=1,N2M1
00930 T1=TOPH(X(K))
00940 T2=TOPH(X(K+1))
00950 U1=T1+HA*Y(I)
00960 U2=T1+HA*Y(I+1)
00970 U3=T2+HA*Y(I)
00980 U4=T2+HA*Y(I+1)
00990 ZL=MINIF(U1,U2,U3,U4)
01000 ZH=MAXIF(U1,U2,U3,U4)
01010 SP=ZH-ZL
01020 TP=PP(K)*PP(I)
01030 IF(SP)15,80,20
01040 15 PRINT 1500
01050 1500 FORMAT("INTERVAL NEGATIVE")
01060 DO 10 200
01070 DO 30 JL=2,M
01080 IF(Z(JL)-ZL)30,30,25
01090 25 ZLOW=ZL
01100 JTEMP=JL
01110 GO TO 40
01120 30 CONTINUE
01130 40 CONTINUE
01140 DO 60 J=JTEMP,M
01150 IF(Z(J)-ZH)50,45,45
01160 45 JSAVE=J
01170 ZHIGH=ZH
01180 GO TO 70
01190 50 ZHIGH=Z(J)
01200 RP(J-1)=KP(J-1)+((ZH-ZLOW)/SP)*TPK
01210 ZLOW=Z(J)
01220

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20 DO 30 JL=2,M
  IF(Z(JL)-ZL)30,30,25
25 ZLOW=ZL
  -- JTEMP=JL
  GO TO 40
30 CONTINUE
40 CONTINUE
  DO 60 J=JTEMP,M
  IF(Z(J)-ZH)50,45,45
45 JSAVE=J
  ZHIGH=ZH
  GO TO 70
50 ZHIGH=Z(J)
  KP(J-1)=KP(J-1)+((ZHIGH-ZLOW)/SP)*TPK
  ZLOW=Z(J)
60 CONTINUE
70 KP(JSAVE-1)=KP(JSAVE-1)+((ZL-GH-ZLOW)/SP)*TPK
  GO TO 100
80 DO 90 J=2,M
  IF(ZL-Z(J))85,82,85
82 KP(J-1)=KP(J-1)+TPK
  JSAVE=J
  GO TO 100
85 IF(ZL-Z(J))82,90,90
90 CONTINUE
  KP(M-1)=KP(M-1)+TPK
  JSAVE=M
100 CONTINUE
120 CONTINUE
200 RETURN
  END
  FUNCTION TOPH(ZZ)
  COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
  COMMON A(20),B(20),C(20),NN,YMEAN
  DIMENSION POL(20)
  POL(1)=1.
  POL(2)=ZZ-A(1)
  *M=NN+1
  DO 20 I=3,M
  POL(I)=ZZ*POL(I-1)-A(I-1)*POL(I-1)-B(I-1)*POL(I-2)
  TOPH=YMEAN
  DO 25 K=1,NN
  TOPH=TOPH+C(K)*POL(K+1)
  RETURN
  END

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OLDOUT FRAME 3

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00000 SUBROUTINE CURFIT(NTYPE,T10,T20,T30,T40)
00010 COMMON A1(A),R1(A),C1(A),A1(A),B1(A),C1(A),D1(A)
00020 COMMON A2(A),R2(A),C2(A),A2(A),B2(A),C2(A),D2(A)
00030 COMMON A3(A),R3(A),C3(A),A3(A),B3(A),C3(A),D3(A)
00040 COMMON A4(A),R4(A),C4(A),A4(A),B4(A),C4(A),D4(A)
00050 COMMON A5(A),R5(A),C5(A),A5(A),B5(A),C5(A),D5(A)
00060 COMMON A6(A),R6(A),C6(A),A6(A),B6(A),C6(A),D6(A)
00070 CALL COEFF
00080 NV=5
00090 NN=4
00100 TS=SORTF(T)
00110 TID=EXPFCOPOLY(TS,A1,R1,C1,A1,B1,C1,A1,MM,NN,VM(1))
00120 T2D=EXPFCOPOLY(TS,A2,R2,C2,A2,B2,C2,A2,MM,NN,VM(2))*1.E3
00130 T3D=EXPFCOPOLY(TS,A3,R3,C3,A3,B3,C3,A3,MM,NN,VM(3))
00140 T4D=EXPFCOPOLY(TS,A4,R4,C4,A4,B4,C4,A4,MM,NN,VM(4))
00150 T5D=EXPFCOPOLY(TS,A5,R5,C5,A5,B5,C5,A5,MM,NN,VM(5))*1.E3
00160 T6D=EXPFCOPOLY(TS,A6,R6,C6,A6,B6,C6,A6,MM,NN,VM(6))*1.E3
00170 GO TO 40
00180 10 T10=EXPFCOPOLY(TS,A1,R1,C1,A1,B1,C1,A1,MM,NN,VM(1))
00190 GO TO 20
00200 15 T20=EXPFCOPOLY(TS,A2,R2,C2,A2,B2,C2,A2,MM,NN,VM(2))
00210 GO TO 30
00220 30 T30=EXPFCOPOLY(TS,A3,R3,C3,A3,B3,C3,A3,MM,NN,VM(3))
00230 GO TO 40
00240 35 T40=EXPFCOPOLY(TS,A4,R4,C4,A4,B4,C4,A4,MM,NN,VM(4))
00250 40 RETURN
00260 END
00270 SUBROUTINE COEFF
00280 COMMON A1(A),R1(A),C1(A),A2(A),R2(A),C2(A),A3(A),R3(A),C3(A)
00290 COMMON A4(A),R4(A),C4(A),A5(A),R5(A),C5(A),A6(A),R6(A),C6(A)
00300 COMMON VM(A)
00310 COMMON DIMMY(1000)
00320 A1(1)=4.978734
00330 A1(2)=11.2840
00340 A1(3)=9.779519
00350 A1(4)=9.67530
00360 R1(1)=R2(1)=R3(1)=R4(1)=R5(1)=R6(1)=0.
00370 R1(2)=41.13013
00380 R1(3)=18.10669
00390 R1(4)=25.06573
00400 C1(1)=6.636589
00410 C1(2)=-.0182150
00420 C1(3)=-.001990154
00430 C1(4)=-.2700020E-3
00440 A2(1)=A2(2)=A2(3)=A2(4)=175.
00450 R2(1)=13125.
00460 R2(2)=10000.
00470 R2(3)=8839.284
00480 C2(1)=-.01707143
00490 C2(2)=-.3183333E-3
00500 C2(3)=5.59594E-7
00510 C2(4)=3.924243E-9
00520 A3(1)=.4870088
00530 A3(2)=11.11457
00540 A3(3)=8.702233
00550 A3(4)=8.578034
00560 R3(1)=4.800994
00570 R3(2)=8.698760
00580 R3(3)=9.727467
00590 C3(1)=-.597208
00600 C3(2)=-5.188701E-3
00610 C3(3)=-9.808954E-4
00620 C3(4)=3.701192E-4
00630 A4(1)=9.492003
00640 A4(2)=14.30388
00650 A4(3)=10.24751
00660 A4(4)=10.63706
00670 R4(1)=12.26415
00680 R4(2)=12.82067
00690 R4(3)=19.59597
00700 C4(1)=-.4797487
00710 C4(2)=-1.475841E-2
00720 C4(3)=1.445613E-4
00730 C4(4)=1.624495E-4

```

EMUL PROGRAM

IBCMAR

1/16/67

```
00000      COMMON PMARG(100),PTOP(100)
00010      COMMON INT(100)
00030      1 PRINT 1000
00040      READ:NMAX,MMAX,T,SUM
00050      READ:(PTOP(I),I=1,MMAX)
00060      READ:(INT(I),I=1,MMAX)
00070      NMAXP=NMAX+1
00110      DO 15 I=1,MMAX
00120      15 PTOP(I)=PTOP(I)/SUM
00130      DO 20 I=1,NMAXP
00140      20 PMARG(I)=0.
00150      RAT=T/(1.-T)
00160      DO 40 K=1,MMAX
00170      N=INT(K)
00180      FN=N
00190      X=0.
00200      PR=(1.-T)**N
00210      PMARG(1)=PMARG(1)+PR*PTOP(K)
00220      25 IF(X-FN)30,40,30
00230      30 PR=PR*(FN-X)*RAT/(X+1.)
00240      X=X+1.
00250      NX=X+1
00260      PMARG(NX)=PMARG(NX)+PR*PTOP(K)
00270      GO TO 25
00280      40 CONTINUE
00290      PRINT 1000
00300      1000 FORMAT(1H1)
00310      SS=0.
00320      DO 50 I=1,NMAXP
00330      NI=I-1
00340      PRINT:NI,PMARG(I)
00350      SS=SS+PMARG(I)
00360      50 CONTINUE
00370      PRINT 1000
00380      PRINT:SS
00390      GO TO 1
00400      END
```

MIXED
1/16/67

SLIST

```
00000 COMMON X(100),Y(100),Z(100),RP(100),M
00010 1 READ:M,(Z(I),I=1,M)
00020 2 READ:ZL,ZH,TPR
00030 CALL ADDER(ZL,ZH,TPR)
00040 PRINT:(RP(I),I=1,M)
00050 PRINT 1000
00060 1000 FORMAT(1H1)
00070 GO TO 2
00080 END
00090 SUBROUTINE ADDER(ZL,ZH,TPR)
00100 COMMON X(100),Y(100),Z(100),RP(100),M
00110 DO 10 I=2,M
00120 10 RP(I-1)=0.
00130 SP=ZH-ZL
00140 IF(SP) 15,80,20
00150 15 PRINT 1500
00160 1500 FORMAT("INTERVAL NEGATIVE")
00170 GO TO 200
00180 20 DO 30 JL=2,M
00190 IF(Z(JL)-ZL) 30,30,25
00200 25 ZLOW=ZL
00210 JTEMP=JL
00220 GO TO 40
00230 30 CONTINUE
00240 40 CONTINUE
00250 DO 60 J=JTEMP,M
00260 IF(Z(J)-ZH) 50,45,45
00270 45 JSAVE=J
00280 ZHIGH=ZH
00290 GO TO 70
00300 50 ZHIGH=Z(J)
00310 RP(J-1)=RP(J-1)+((ZHIGH-ZLOW)/SP)*TPR
00320 ZLOW=Z(J)
00330 60 CONTINUE
00340 70 RP(JSAVE-1)=RP(JSAVE-1)+((ZHIGH-ZLOW)/SP)*TPR
00350 GO TO 100
00360 80 DO 90 J=2,M
00370 IF(ZL-Z(J)) 85,82,85
00380 82 RP(J-1)=RP(J-1)+TPR
00390 GO TO 100
00400 85 IF(ZL-Z(J)) 82,90,90
00410 90 CONTINUE
00415 RP(M-1)=RP(M-1)+TPR
00420 100 CONTINUE
00430 120 CONTINUE
00440 200 RETURN
00450 END
```

OLD PROGRAM NAME--HELIO3
WAIT.

HELIO 3
1/16/67

READY.

*LIST

```
00000 COMMON DIMMY(78)
00010 COMMON X1(100),X2(100),X3(100),X4(100),7(100)
00020 COMMON P1(100),P2(100),P3(100),P4(100),RP(100)
00030 COMMON XC2(100),XC3(100)
00040 PRINT 1000
00050 RAD=57.2957795
00060 999 FORMAT('1 DAYS TO IMPACT, ORBIT TYPE,T,R YARS "')
00070 1000 FORMAT(' HELIOCENTRIC ORBIT PROBABILITY PROGRAM')
00080 READ:TIME,NTYPE,TP,RRP
00090 CALL CUFFIT(TIME,NTYPE,C1,C2,C3,C4)
00100 PRINT 999
00110 998 FORMAT(' NUMBER, VELOCITY MAG VALUES(M/S)')
00120 READ:N1,(X1(I),I=1,N1)
00130 N1M1=N1-1
00140 PRINT 997
00150 997 FORMAT(' VELOCITY MAG PROBABILITIES')
00160 READ:(P1(I),I=1,N1M1)
00170 PRINT 994
00180 996 FORMAT(' NUMBER,POLAR ANGLE VALUES(DEG)')
00190 READ:N2,(X2(I),I=1,N2)
00200 PRINT 1010
00210 1010 FORMAT(' POLAR ANGLE PROBABILITIES')
00220 READ:N2=N2-1
00230 READ:(P2(I),I=1,N2M1)
00240 PRINT 1015
00250 1015 FORMAT(' NUMBER, N-T PLANE ANGLE VALUES(DEG)')
00260 READ:N3,(X3(I),I=1,N3)
00270 N3M1=N3-1
00280 PRINT 1020
00290 1020 FORMAT(' N-T PLANE ANGLE PROBABILITIES')
00300 READ:(P3(I),I=1,N3M1)
00310 PRINT 1025
00320 1025 FORMAT(' NUMBER, M/CDA VALUES(SLUGS/FT*FT)')
00330 READ:N4,(X4(I),I=1,N4)
00340 N4M1=N4-1
00350 PRINT 1027
00360 1027 FORMAT(' M/CDA PROBABILITIES')
00370 READ:(P4(I),I=1,N4M1)
00380 DO 5000 I=1,N2
00385 X2(I)=SINF(00)
00390 XC2(I)=COSF(00)
00400 DO 5010 I=1,N3
00405 X3(I)=SINF(00)
00410 XC3(I)=COSF(00)
00420 PRINT 1030
00430 1030 FORMAT(' NUMBER, MISS DISTANCE VALUES (KM)')
00440 READ:M,(Z(I),I=1,M)
00450 PRINT 1030
00460 1030 FORMAT(' T1/DVT T2/DVN T3/M/CDA R/DVR
+ C-FU')
00470 PRINT:C1,C2,C3,C4
00480 CALL CROSS(N1M1,N2M1,N3M1,N4M1,C1,C2,C3,C4,TP,RRP,
+JSAVE,M)
00490 MT=M
00500 5 MT=MT-1
```

FOI/DOH FRAME


```

00450 PRINT 1000
00460 FORMAT('9', T1/DVT, T2/DVN, T3CM/CDA), RZ/DVR
00470 + C-F")
00480 PRINT: C1, C2, C3, C4
00490 CALL CROSS(N1M1, N2M1, N3M1, N4M1, V4M1, C1, C2, C3, C4, TP, RRP,
00500 +JSAVE, M)
00510 MT=M
00520 5 MT=MT-1
00530 IF(RP(MT)) 6, 6, 7
00540 6 IF(MT-1) 7, 7, 5
00550 7 JM1=MT
00560 T=0.
00570 DO 10 I=1, JM1
00580 T=T+RP(I)
00590 10 PRINT 1060
00600 FORMAT('9', +++) MISS DISTANCE PPARADILITIES +++)
00610 PRINT: RP(1), I=1, JM1)
00620 PRINT 1070, T
00630 FORMAT('CHECK SUM =', F10.6)
00640 PRINT 1001
00650 FORMAT(IH1)
00660 GO TO 9
00670 FND
00680 SUBROUTINE CROSS(N1M1, N2M1, N3M1, N4M1, C1, C2, C3, C4,
00690 +TP, RRP, JSAVE, M)
00700 COMMON DIMY(78)
00710 COMMON X1(100), X2(100), X3(100), X4(100), Z(100)
00720 COMMON P1(100), P2(100), P3(100), P4(100), RP(100)
00730 COMMON X00(100), X03(100)
00740 DIMENSION O(20)
00750 DO 10 I=0, M
00760 10 RP(I-1)=0.
00770 DO 100 I1=1, N1M1
00780 DO 100 I2=1, N2M1
00790 DO 100 I3=1, N3M1
00800 DO 100 I4=1, N4M1
00810 TP=RP(I1)*P2(I2)*P3(I3)*P4(I4)
00820 JZ=0
00830 DO 20 I01=1, 2
00840 I71=I1+I01-1
00850 DO 30 I02=1, 2
00860 I72=I2+I02-1
00870 DO 40 I03=1, 2
00880 I73=I3+I03-1
00890 DO 50 I04=1, 2
00900 I74=I4+I04-1
00910 JZ=JZ+1
00920 O(I72)=ZZZ(C1, C2, C3, C4, X1(I71), X2(I72), X3(I73), X4(I74),
00930 +X00(I72), X03(I73), TP, RRP)
00940 50 CONTINUE
00950 40 CONTINUE
00960 30 CONTINUE
00970 20 CONTINUE
00980 ZL=MIN(F(O(1), O(2), O(3), O(4), O(5), O(6), O(7), O(8), O(9),
00990 +O(10), O(11), O(12), O(13), O(14), O(15), O(16))
01000 ZH=MAX(F(O(1), O(2), O(3), O(4), O(5), O(6), O(7), O(8), O(9),
01010 +O(10), O(11), O(12), O(13), O(14), O(15), O(16))
01020 SP=ZH-ZL
01030 CALL ODDD(SP, TPR, ZL, ZH, JSAVE, M)
01040 100 CONTINUE
01050 110 CONTINUE
01060 120 CONTINUE
01070 130 CONTINUE
01080 RETURN
01090 FND
01100 FUNCTION ZZ7(O1, D2, O3, D4, VM, STM, SPM, RT, CTV, CPX, TP, RRP)
01110 T=VM*STM*(O1*SPM+D2*CPX)+D3/RT
01120 R=D4*VM*CTM
01130 ZZ7=SORTF((T-TP)**2+(R-RRP)**2)
01140 RETURN
01150 FND

```

FOLDOUT FRAME 2

LIST

```

00006 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
00010 COMMON A(20),B(20),NN,C
00020 PRINT 1000
00030 PRINT 999
00040 FORMAT("1 ORBIT TYPE(1-6)")
00050 FORMAT(" PERIAPSIS DISTRIBUTION PROGRAM")
00060 READ:11
00070 CALL OTYPE(I1,MM)
00080 A(I)=A(I)/2.
00090 A(NN)=A(NN)/2.
00100 FFM=MM
00110 C=3.14159265/FFM
00120 PRINT 1010
00130 FORMAT("1 NUMBERS, ANGLE ATTACK VALUES")
00140 READ:N1,(X(I),I=1,N1)
00150 N1=N1-1
00160 PRINT 1015
00170 FORMAT(" ANGLE PROBABILITIES, 90. TO 270.")
00180 READ:(P(I),I=1,N1M1)
00190 PRINT 1020
00200 FORMAT(" NUMBERS, VALUES FOR PERIAPSIS DECREMENT")
00210 READ:M,(Z(I),I=1,M)
00220 PRINT 1030
00230 FORMAT(" NUMBERS, VALUES FOR VELOCITY INCREMENT")
00240 READ:N2,(Y(I),I=1,N2)
00250 PRINT 1040
00260 FORMAT(" VELOCITY INCREMENT PROBABILITIES")
00270 N2=N2-1
00280 READ:(PP(I),I=1,N2M1)
00290 CALL ADDER(N1M1,N2M1,M,JSAVE)
00300 MT=M
00310 N1=N1-1
00320 IF(NP(M1))6,6,7
00330 I=I+NP(I)
00340 I=I+NP(I)
00350 DO 10 I=1,N1
00360 T=0.
00370 PRINT 1060
00380 PRINT 1070
00390 PRINT 1080
00400 PRINT 1090
00410 GO TO 2
00420 END
00430 SUBROUTINE ADDER(N1M1,N2M1,M,JSAVE)
00440 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
00450 DO 10 I=2,M
00460 RP(I-1)=0.
00470 DO 100 K=1,N1M1
00480 DO 100 I=1,N2M1
00490 T1=PDV(X(K))
00500 T2=PDV(X(K+1))
00510 U1=T1+Y(I)
00520 U2=T1+Y(I+1)

```

FORGOTTEN

```

00470 DO 100 N=1,NM1
00480 DO 100 J=1,N2M1
00482 I1=PDV(X(K))
00483 I2=PDV(X(K+1))
00490 U1=I1*Y(I)
00500 U2=I1*Y(I+1)
00510 U3=I2*Y(I)
00520 U4=I2*Y(I+1)
00530 ZL=MIN1F(U1,U2,U3,U4)
00540 ZH=MAX1F(U1,U2,U3,U4)
00550 SP=ZH-ZL
00560 IPR=P(K)*PF(I)
00570 IF(SP)15,80,20
00580 15 PRINT 1500
00590 1500 FORM1("INTERVAL NEGATIVE")
00600 GO TO 200
00610 DO 30 JL=2,M
00620 IF(Z(JL)-ZL)30,30,25
00630 25 ZLOW=ZL
00640 JTEMP=JL
00650 GO TO 40
00660 30 CONTINUE
00670 40 CONTINUE
00680 DO 60 J=JTEMP,N
00690 IF(Z(J)-ZH)50,45,45
00700 JSAVE=J
00710 ZHIGH=ZH
00720 GO TO 70
00730 50 ZHIGH=Z(J)
00740 KP(J-1)=RP(J-1)+((ZHIGH-ZLOW)/SP)*TPR
00750 ZLOW=Z(J)
00760 60 CONTINUE
00770 70 KP(JSAVE-1)=RP(JSAVE-1)+((ZHIGH-ZLOW)/SP)*TPR
00780 GO TO 100
00790 80 DO 90 J=2,M
00800 IF(ZL-Z(J-1))85,82,85
00810 82 KP(J-1)=RP(J-1)+TPR
00820 JSAVE=J
00830 GO TO 100
00840 85 IF(ZL-Z(J))82,90,90
00850 90 CONTINUE
00860 KP(M-1)=RP(M-1)+TPR
00870 JSAVE=M
00880 100 CONTINUE
00890 120 CONTINUE
00900 200 RETURN
00910 END
00920 FUNCTION PDV(ZZ)
00930 COMMON X(100),Y(100),Z(100),P(100),PF(100),RP(100)
00940 COMMON A(20),B(20),NN,C
00950 ZZ=(ZZ-90.)/10.
00960 PDV=0.
00970 DO 15 J=1,NN
00980 IJ=J-1
00990 G=C*FJ*ZZ
01000 15 PDV=PDV+A(J)*COSF(G)+B(J)*SINF(G)
01010 RETURN
01020 END

```

FOLDS 2

WAIT.

DELP2
1/16/67

READY.

SLIST

```

00000 C
00010 CURVE FITTING PROGRAM FOR DELP1
00020 SUBROUTINE CTYPE(IT,MM)
00030 COMMON DUMMY(6000)
00040 COMMON A(20),R(20),NN,C
00050 NN=10
00060 MM=9
00070 DO 20 I=1,20
00080 R(I)=0.
00090 GO TO (100,200,300,400,500,600),IT
00100 A(1)=10.59444
00110 A(2)=-3.348694
00120 A(3)=-.7277126
00130 A(4)=-.2944445
00140 A(5)=-.179111
00150 A(6)=-.1401124
00160 A(7)=-.1388889
00170 A(8)=-.09452752
00180 A(9)=-.07650987
00190 A(10)=-.09444452
00200 RETURN
00210 A(1)=9.794444
00220 A(2)=-3.100466
00230 A(3)=-.6714439
00240 A(4)=-.2611111
00250 A(5)=-.1491711
00260 A(6)=-.1161407
00270 A(7)=-.1055556
00280 A(8)=-.1000602
00290 A(9)=-.09605177
00300 A(10)=-.09444452
00310 RETURN
00320 A(1)=9.194444
00330 A(2)=-2.9040966
00340 A(3)=-.6530806
00350 A(4)=-.2722222
00360 A(5)=-.1060668
00370 A(6)=-.1198379

```

FORGET FRAME

00280	A(9)=-.09605177	
00290	A(10)=-.09444452	
00300	RETURN	
00310	A(1)=9.194444	300
00320	A(2)=-2.9040966	
00330	A(3)=-.6530806	
00340	A(4)=-.2722222	
00350	A(5)=-.1060668	
00360	A(6)=-.1198379	
00370	A(7)=-.07222224	
00380	A(8)=-.05586342	
00390	A(9)=-.09085276	
00400	A(10)=-.07222228	
00410	RETURN	
00430	A(1)=15.69444	400
00440	A(2)=-5.006866	
00450	A(3)=-1.155144	
00460	A(4)=-.4277778	
00470	A(5)=-.3001248	
00480	A(6)=-.2253168	
00490	A(7)=-.1388889	
00500	A(8)=-.1511513	
00510	A(9)=-.1280644	
00520	A(10)=-.1277779	
00530	RETURN	
00540	A(1)=15.03889	500
00550	A(2)=-4.891183	
00560	A(3)=-1.157197	
00570	A(4)=-.45	
00580	A(5)=-.3137546	
00590	A(6)=-.1722799	
00600	A(7)=-.09444444	
00610	A(8)=-.08653781	
00620	A(9)=-.07904804	
00630	A(10)=-.05000005	
00640	RETURN	
00650	A(1)=14.39444	600
00660	A(2)=-4.944771	
00670	A(3)=-1.200388	
00680	A(4)=-.4944444	
00690	A(5)=-.2284768	
00700	A(6)=-.1095758	
00710	A(7)=-.07222217	
00720	A(8)=-.03798681	
00730	A(9)=-.07886461	
00740	A(10)=-.1055556	
00745	RETURN	
00750	FND	

FOLDOUT FRAME 2

READY.

HELIO1
1/16/67

LIST

```

00000 COMMON DUMMY(78)
00010 COMMON X1(100),X2(100),X3(100),X4(100),Z(100)
00020 COMMON P1(100),P2(100),P3(100),P4(100),RP(100)
00030 PRINT 1000
00040 2 PRINT 999
00050 999 FORMAT("1 DAYS TO IMPACT, ORBIT TYPE,T,R MARS ")
00060 1000 FORMAT(" HELIOCENTRIC ORBIT PROBABILITY PROGRAM")
00070 READ:TIME,NTYPE,TP,RRP
00080 CALL CUFIT(TIME,NTYPE,C1,C2,C3,C4)
00090 PRINT 998
00100 998 FORMAT(" NUMBER, TANGENTIAL VELOCITY VALUES")
00110 READ:N1,(X1(I),I=1,N1)
00120
00130 N1M1=N1-1
00140 PRINT 997
00150 997 FORMAT(" TANGENTIAL PROBABILITIES")
00160 READ:(P1(I),I=1,N1M1)
00170 PRINT 996
00180 996 FORMAT(" NUMBER, NORMAL VELOCITY VALUES")
00190 READ:N2,(X2(I),I=1,N2)
00200 PRINT 1010
00210 1010 FORMAT(" NORMAL PROBABILITIES")
00220 N2M1=N2-1
00230 READ:(P2(I),I=1,N2M1)
00240 PRINT 1015
00250 1015 FORMAT(" NUMBER, M/CDA VALUES")
00260 READ:N3,(X3(I),I=1,N3)
00270 N3M1=N3-1
00280 PRINT 1020
00290 1020 FORMAT(" M/CDA PROBABILITIES")
00300 READ:(P3(I),I=1,N3M1)
00310 PRINT 1025
00320 1025 FORMAT(" NUMBER, O-O-P VELOCITY VALUES")
00330 READ:N4,(X4(I),I=1,N4)
00340 N4M1=N4-1
00350 PRINT 1027
00360 1027 FORMAT(" O-O-L PROBABILITIES")
00370 READ:(P4(I),I=1,N4M1)
00380 PRINT 1030
00390 1030 FORMAT(" NUMBER, MISS DISTANCE VALUES")
00400 READ:M,(Z(I),I=1,M)
00410 PRINT 1080
00420 1080 FORMAT("0 T1/DVT T2/DWN T3(M/CDA) R/DVR C-F")
00430 PRINT:C1,C2,C3,C4
00440 CALL GROSS(N1M1,N2M1,N3M1,N4M1,C1,C2,C3,C4,TP,RRP,
00450 +JSAVE,M)
00460 MT=M
00470 5 MT=MT-1
00480 IF(RP(MT))6,6,7
00490 6 IF(MT-1)7,7,5
00500 7 JM1=MT
00510 T=0.
00520 DO 10 I=1,JM1
00530 10 T=T+RP(I)
00540 PRINT 1060
00550 1060 FORMAT(" IN-PLANE MISS DISTANCE PROBABILITIES")
005 0 PRINT:(RP(I),I=1, JM1)

```

NO. 1000 TRANS

```

00470
00480 5 MT=MT-1
00490 6 IF(RP(MT)) 6, 6, 7
00500 7 JM1=MT
00510 T=0.
00520 DO 10 I=1, JM1
00530 T=T+RP(I)
00540 PRINT 1060
00550 1060 FORMAT(" IN-PLANE MISS DISTANCE PROBABILITIES")
00560 PRINT:(RP(I), I=1, JM1)
00570 PRINT 1070, T
00580 1070 FORMAT("CHECK SUM =", F10.6)
00590 PRINT 1001
00600 1001 FORMAT(:H1)
00610 GO TO 2
00620 END
00630 SUBROUTINE GROSS(N1M1, N2M1, N3M1, N4M1, C1, C2, C3, C4,
00640 +TP, RRP, JSAVE, M)
00650 COMMON DUMMY(78)
00660 COMMON X1(100), X2(100), X3(100), X4(100), Z(100)
00670 COMMON P1(100), P2(100), P3(100), P4(100), RP(100)
00680 DIMENSION O(20)
00690 DO 10 I=2, M
00700 10 RP(I-1)=0.
00710 DO 130 I1=1, N1M1
00720 DO 120 I2=1, N2M1
00730 DO 110 I3=1, N3M1
00740 DO 100 I4=1, N4M1
00750 TPR=P1(I1)*P2(I2)*P3(I3)*P4(I4)
00760 JZ=0
00770 DO 20 I01=1, 2
00780 I21=I1+I01-1.
00790 DO 30 I02=1, 2
00800 I22=I2+I02-1
00810 DO 40 I03=1, 2
00820 I23=I3+I03-1
00830 DO 50 I04=1, 2
00840 I24=I4+I04-1
00850 JZ=JZ+1
00860 O(JZ)=ZZZ(C1, C2, C3, C4, X1(I21), X2(I22), X3(I23), X4(I24)
00870 +, TP, RRP)
00880 50 CONTINUE
00890 40 CONTINUE
00900 30 CONTINUE
00910 20 CONTINUE
00920 ZL=MIN1F(O(1), O(2), O(3), O(4), O(5), O(6), O(7), O(8), O(9),
00930 +O(10), O(11), O(12), O(13), O(14), O(15), O(16))
00940 ZH=MAX1F(O(1), O(2), O(3), O(4), O(5), O(6), O(7), O(8), O(9),
00950 +O(10), O(11), O(12), O(13), O(14), O(15), O(16))
00960 SP=ZH-ZL
00970 80 CALL DDDD(SP, TPR, ZL, ZH, JSAVE, M)
00980 100 CONTINUE
00990 110 CONTINUE
01000 120 CONTINUE
01010 130 CONTINUE
01020 RETURN
01030 END
01040 FUNCTION ZZZ(D1, D2, D3, D4, G1, G2, G3, G4, TP, RRP)
01050 T=G1*D1+G2*D2+G3/D3
01060 R=G4*D4
01070 ZZZ= SORTF((T-TP)**2+(R-RRP)**2)
01080 RETURN
01090 END
01100
01110
01120
01130
01140
01150

```

FOLDOUT FRAME 2

SLIS.

```
00090 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
00010 COMMON A(20),B(20),C(20),NN,YMEAN
00020 PRINT 1000
00030 PRINT 999
00040 999 FORMAT("1 ORBIT TYPE(1-6),ATMOS TYPE(1 OR 2)")
00050 1000 FORMAT(" M/CDA DISTRIBUTION PROGRAM")
00060 NN=4
00070 READ:IT,ITT
00080 A(1)=A(2)=A(3)=A(4)=600.
00090 B(1)=0.
00100 B(2)=66666.67
00110 B(3)=51333.33
00120 B(4)=46285.71
00130 GO TO (60,80),ITT
00140 60 C(1)=-.01961519
00150 C(2)=-4.065703E-5
00160 C(3)=8.695401E-8
00170 C(4)=-1.711093E-10
00180 YMEAN=-.1833621
00190 GO TO 90
00200 80 C(1)=-.0157079
00210 C(2)=-2.496592E-5
00220 C(3)=5.334886E-8
00230 C(4)=-1.370223E-10
00240 YMEAN=-3.913773
00250 90 GO TO (91,92,93,94,95,96),IT
00260 91 HA=10000.
00270 PA=1000.
00280 GO TO 100
00290 92 HA=10000.
00300 PA=500.
00310 GO TO 100
00320 93 HA=10000.
00330 PA=200.
00340 GO TO 100
00350 94 HA=20000.
00360 PA=1000.
00370 GO TO 100
00380 95 HA=20000.
00390 PA=500.
00400 GO TO 100
00410 96 HA=20000.
00420 PA=200.
00430 100 PRINT 998
00440 998 FORMAT(" APOAPSIS(KM) PERIAPSIS(KM)")
00450 PRINT:HA,PA
00460 2 PRINT 1010
00470 1010 FORMAT(" NUMBER, PERIAPSIS VALUES")
00480 READ:N1,(X(I),I=1,N1)
00490 N1=N1-1
00500 PRINT 1015
00510 1015 FORMAT(" PERIAPSIS PROBABILITIES")
00520 READ:(P(I),I=1,N1)
00530 PRINT 1020
00540 1020 FORMAT(" NUMBER,VALUES FOR M/CDA")
00550 READ:M,(Z(I),I=1,M)
00560 PRINT 1030
00570 1030 FORMAT(" NUMBER,VALUES FOR TIME IN ORBIT")
00580 READ:N2,(Y(I),I=1,N2)
00590 PRINT 1040
00600 1040 FORMAT(" TIME IN ORBIT PROBABILITIES")
00610 N2=N2-1
```



```

00560      I=J
00570 1030 FORMAT(" NUMBER,VALUES FOR TIME IN ORBIT")
00580 READ:N2,(Y(I),I=1,N2)
00590 PRINT 1040
00600 1040 FORMAT(" TIME IN ORBIT PROBABILITIES")
00610 N2M1=N2-1
00620 READ:(PP(I),I=1,N2M1)
00630 CALL MULPY(N1M1,N2M1,M,JSVAE,HA)
00640 MT=M
00650 5 MT=MT-1
00660 IF<RP(MT)>6,6,7
00670 6 IF(MT-1)7,7,5
00680 7 JM1=MT
00690 T=0.
00700 DO 10 I=1,JM1

00700 10 T=I+RP(I)
00710 PRINT 1060
00720 1060 FORMAT(" M/CDA DISTRIBUTION")
00730 PRINT:(RP(I),I=1,JM1)
00740 PRINT 1070,T
00750 1070 FORMAT("CHECK SUM =",F10.6)
00760 PRINT 1001
00770 1001 FORMAT(1H1)
00780 GO TO 2
00790 END
00800 SUBROUTINE MULPY(N1M1,N2M1,M,JSVAE,HA)
00810 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
00820 COMMON A(20),B(20),C(20),NN,YMEAN
00830 DO 10 I=2,M
00840 10 RP(I-1)=0.
00850 DO 120 K=1,N1M1
00860 DO 100 I=1,N2M1
00870 T1=TOPH(X(K))
00880 T2=1-IPH(X(K+1))
00890 Q1=Y(I)/(HA*T1)
00900 Q2=Y(I+1)/(HA*T2)
00910 Q3=Y(I)/(HA*T2)
00920 Q4=Y(I+1)/(HA*T2)
00930 ZL=MINIF(Q1,Q2,Q3,Q4)
00940 ZH=MAXIF(Q1,Q2,Q3,Q4)
00950 SP=ZH-ZL
00960 TPR=P(K)+PP(I)
00970 IF(SP)15,80,20

00980 15 PRINT 1500

00990 1500 FORMAT("INTERVAL NEGATIVE")
01000 GO TO 200
01010 20 DO 30 JL=2,M
01020 IF(Z(JL)-ZL)30,30,25
01030 25 ZLOW=ZL
01040 JTEMP=JL
01050 GO TO 40
01060 30 CONTINUE
01070 40 CONTINUE
01080 DO 60 J=JTEMP,M
01090 IF(Z(J)-ZH)50,45,45
01100 45 JSVAE=J
01110 ZHIGH=ZH
01120 GO TO 70
01130 50 ZHIGH=Z(J)
01140 RP(J-1)=RP(J-1)+((ZHIGH-ZLOW)/SP)*TPR
01150 ZLOW=Z(J)
01160 60 CONTINUE
01170 70 RP(JSVAE-1)=RP(JSVAE-1)+((ZHIGH-ZLOW)/SP)*TPR
01180 GO TO 100

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 202004 11:11 AM

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01050 25 ZLOW=ZL
01060 JTEMP=JL
01070 GO TO 40
01080 9 CONTINUE
01090 40 CONTINUE
01100 DO 60 J=JTEMP,M
01110 IF(Z(J)-ZH)50,45,45
01120 45 JSAVE=J
01130 ZHIGH=ZH
01140 GO TO 70
01150 50 ZHIGH=Z(J)
01160 RP(J-1)=RP(J-1)+((ZHIG-H-ZLOW)/SP)*TPR
01170 ZLOW=Z(J)
01180 60 CONTINUE
01190 70 RP(JSAVE-1)=RP(JSAVE-1)+((ZHIG-H-ZLOW)/SP)*TPR
01200 GO TO 100
01210 80 DO 90 J=2,M
01220 IF(ZL-Z(J))85,82,85
01230 82 RP(J-1)=RP(J-1)+TPR
01240 JSAVE=J
01250 GO TO 100
01260 85 IF(ZL-Z(J))82,90,90
01270 90 CONTINUE
01280 RP(M-1)=RP(M-1)+TPR
01290 JSAVE=M
01300 100 CONTINUE
01310 120 CONTINUE
01320 200 RETURN
01330 END
01340 FUNCTION TOPH(ZZ)
01350 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
01360 DIMENSION POL(20)
01370 POL(1)=1.
01380 POL(2)=ZZ-A(1)
01390 MM=NN+1
01400 DO 20 I=3,MM
01410 20 POL(I)=ZZ+POL(I-1)-A(I-1)+POL(I-1)-B(I-1)+POL(I-2)
01420 TOPH=YMEAN
01430 DO 25 K=1,NN
01440 25 TOPH=TOPH+C(K)+POL(K+1)
01450 RETURN
01460 END

```

SOLOUT FRAM 3

14

REA

AGRID
1/16/67

SLIST

```
00000 C234567
00010 C   GRID - ROUTINE TO CALCULATE NEW PROB. TO FIT NORMAL GRID
00020     FUNCTION GRID(M,PROB,Q,A,B)
00030     DIMENSION Q(1),PROB(1)
00040     GRID = 0.
00050     DO 10 I=1,M
00060     IF(Q(I)-A) 10,10,5
00070     5  IS = I
00080     GO TO 20
00090     10 CONTINUE
00100     20 ATEMP = A
00110     DO 50 K=IS,M
00120     IF(Q(K)-B) 25,25,23
00130     23 KT = K
00140     GO TO 100
00150     25 GRID = GRID+((Q(K)-ATEMP)/(Q(K)-Q(K-1)))*PROB(K-1)
00160     ATEMP = Q(K)
00170     50 CONTINUE
00180     60 GO TO 500
00190     100 GRID = GRID+((B-ATEMP)/(Q(KT)-Q(KT-1)))*PROB(KT-1)
00200     500 RETURN
00210     END
00220 C234567
00230 C   MAR - ROUTINE TO CALCULATE MARGINAL PROBABILITIES
00240     SUBROUTINE MAR(MMAX,T,PTOP,INT,PMARG)
00250     DIMENSION PTOP(1),INT(1),PMARG(1)
00260     NMAXP = 101
00270     DO 20 I=1,NMAXP
00280     20 PMARG(I) = 0.0
00290     RAT = T/(1.-T)
00300     DO 40 K=1,MMAX
00310     N = INT(K)
00320     FN = N
00330     X = 0.0
00340     PR = (1.-T)**N
00350     PMARG(1) = PMARG(1)+PR*PTOP(K)
00360     25 IF(X-FN) 30,40,30
00370     30 PR = PR*(FN-X)*RAT/(X+1.)
00380     X = X+1.
00390     NX = X+1
00400     PMARG(NX) = PMARG(NX)+PR*PTOP(K)
00410     GO TO 25
00420     40 CONTINUE
00430     RETURN
00440     END
```

```

000006 C *****UPPER LIMIT PROGRAM*****
000010 COMMON ALP(100),ULP(100),Z(100),P(100),ZOUT(100)
000020 COMMON POUT(100),PT(100),PF(100),ZZ(100)
000030 C FORMAT STATEMENTS*****
000040 1000 FORMAT("1 PROGRAM TO COMBINE UPPER LIMIT AND"
000050 + " M/GDA VARIABLES")
000060 1020 FORMAT("0 NUMBER,UPPER LIMIT POINTS")
000070 1040 FORMAT(" UPPER LIMIT PROBABILITIES")
000080 1060 FORMAT(" NUMBER,M/GDA POINTS")
000090 1080 FORMAT(" M/GDA PROBABILITIES")
000100 2000 FORMAT(" NUMBER,OUTPUT POINTS")
000110 2020 FORMAT("0 FIRST LAST PROBABILITY")
000120 C*****
000130 PRINT 1000
000140 1 CONTINUE
000150 PRINT 1020
000160 READ:NUL,(ALP(K),K=1,NUL)
000170 NLM1=NUL-1
000180 PRINT 1040
000190 READ:(ULP(K),K=1,NLM1)
000200 PRINT 1060
000210 READ:N,(Z(I),I=1,N)
000220 NM1=N-1
000230 PRINT 1080
000240 READ:(P(I),I=1,NM1)
000250 PRINT 2000
000260 READ:NOUT,(ZOUT(I),I=1,NOUT)
000270 NM1=NOUT-1
000280 DO 3 I=1,NM1
000290 3 POUT(I)=0.
000300 C*****
000310 DO 200 K=2,NUL
000320 SUM=0.
000330 ZZ(1)=Z(1)
000340 IF(Z(1)-ALP(K))8,200,200
000350 8 DO 50 J=2,N
000360 IF(Z(J)-ALP(K))20,10,10
000370 10 ZZ(J)=ALP(K)
000390 DEN=Z(J)-Z(J-1)
000400 IF(DEN)15,12,15
000410 12 PP(J-1)=P(J-1)
000420 10 10 18
000430 15 PP(J-1)=P(J-1)*((ALP(K)-Z(J-1))/DEN)
000440 18 N1=J
000450 NTM1=N1-1
000455 SUM=SUM+PP(J-1)
000460 GO TO 60
000470 20 PP(J-1)=P(J-1)
000480 ZZ(J)=Z(J)
000490 SUM=SUM+PP(J-1)
000500 50 CONTINUE
000510 ZZ(N+1)=ALP(K)
000520 NT=N+1
000530 NTM1=N
000540 PP(N)=0.
000550 C*****
000560 60 DO 70 I=1,NM1
000570 70 PP(I)=PP(I)+1

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REVERSE TRANSMIT

```

00490 SUM=SUM+PP(J-1)
00500 CONTINUE
00510 ZZ(N+1)=ALP(K)
00520 NT=N+1
00530 NTI=N
00540 PP(N)=0.
00550 C*****
00560 DO 70 I=1,NTMI
00570 PP(I)=PP(1)/SUM
00580 DO 80 I=1,NOMI
00590 GTP(I)=0.
00600 DO 90 I=1,NTMI
00610 SP=ZZ(I+1)-ZZ(I)
00620 CALL LOAD(SP,PP(I),ZZ(I),ZZ(I+1),NOUT,ZOUT,GTP)
00630 DO 100 I=1,NOMI
00640 POUT(I)=POUT(1)+GTP(I)*ULP(K-1)
00650 CONTINUE
00660 C*****
00670 PRINT 2020
00680 DO 230 I=1,NOMI
00690 PRINT:ZOUT(I),ZOUT(I+1),POUT(I)
00700 GO TO 1
00710 END
00720 C*****
00730 SUBROUTINE LOAD(SP,TPR,ZL,ZHM,Y,NP)
00740 DIMENSION Y(1),NP(1)
00750 IF(SP)GO,80,20
00760 DO 30 JL=2,0
00770 IF(Y(JL)-ZL)30,30,25
00780 ZL:=ZL
00790 JTMP=JL
00800 GO TO 40
00810 CONTINUE
00820 DO 60 J=JTMP,M
00830 IF(Y(J)-ZM)50,45,45
00840 JSAVE=J
00850 ZHIGH=ZM
00860 GO TO 70
00870 ZHIGH=Y(J)
00880 KP(J-1)=KP(J-1)+((ZHIGH-ZLOW)/SP)*TPR
00890 ZLOW=Y(J)
00900 CONTINUE
00910 KP(JSAVE-1)=KP(JSAVE-1)+((ZHIGH-ZLOW)/SP)*TPR
00920 GO TO 100
00930 DO 90 J=2,M
00940 IF(ZL-Y(J-1))85,82,85
00950 KP(J-1)=KP(J-1)+TPR
00960 JSAVE=J
00970 GO TO 100
00980 IF(ZL-Y(J))82,90,90
00990 CONTINUE
01000 KP(M-1)=KP(M-1)+TPR
01010 JSAVE=M
01020 CONTINUE
01030 RETURN
01040 END
01050

```

FOLOUT FRAM 2


```

SUBROUTINE ULF(I,IM1,Z,IM,JSAVE,HA)
COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
COMMON A(20),B(20),C(20),NN,YMEAN
DO 10 I=2,M
  RP(I-1)=0
10 RP(I-1)=0
DO 120 K=1,NIM1
DO 100 I=1,N2M1
  T1=TOPH(X(K))
  T2=TOPH(X(K+1))
  Q1=T1*HA*Y(I)
  Q2=T1*HA*Y(I+1)
  Q3=T2*HA*Y(I)
  Q4=T2*HA*Y(I+1)
  ZL=MIN(IF(Q1,Q2,Q3,Q4)
  ZH=MAX(IF(Q1,Q2,Q3,Q4)
  SP=ZH-ZL
  TPR=PP(K)*PP(I)
  IF(SP)15,80,20
15 PRINT 1500
1500 FORMAT("INTERVAL NEGATIVE")
  GO TO 200
20 DO 30 JL=2,M
  IF(Z(JL)-ZL)30,30,25
25 ZLOW=ZL
  JTEMP=JL
  GO TO 40
30 CONTINUE
40 CONTINUE
  DO 60 J=JTEMP,M
  IF(Z(J)-ZH)50,45,45
45 JSAVE=J
  ZHIGH=ZH
  GO TO 70
50 ZHIGH=Z(J)
  RP(J-1)=RP(J-1)+((ZHIGH-ZLOW)/SP)*TPR
  ZLOW=Z(J)
60 CONTINUE
70 RP(JSAVE-1)=RP(JSAVE-1)+((ZHIGH-ZLOW)/SP)*TPR
  GO TO 100
80 DO 90 J=2,M
  IF(ZL-Z(J-1))85,82,85
82 RP(J-1)=RP(J-1)+TPR
  JSAVE=J
  GO TO 100
85 IF(ZL-Z(J))82,90,90
90 CONTINUE
  RP(M-1)=RP(M-1)+TPR
  JSAVE=M
100 CONTINUE
120 CONTINUE
200 RETURN
END
FUNCTION TOPH(ZZ)
COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
COMMON A(20),B(20),C(20),NN,YMEAN
DIMENSION POL(20)
POL(1)=1.
POL(P)=ZZ-AK(1)
MM=NN+1
DO 20 I=3,MM
  POL(I)=ZZ*POL(I-1)-A(I-1)*POL(I-1)+POL(I-2)
  TOPH=YMEAN
DO 25 K=1,NN
  TO PH=TOPH+C(K)*POL(K+1)
  TO PH=EXPF(TO PH)
  RETURN
END

```

FOUR PAGE 2

SLIS.

```
00000 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
00010 COMMON A(20),B(20),C(20),NN,YMEAN
00020 PRINT 1000
00030 PRINT 999
00040 FORMAT('1 DEGREE, APOAPSIS, YMEAN')
00050 FORMAT(' M/CDA DISTRIBUTION PROGRAM')
00060 READ:NN,HA,YMEAN
00070 PRINT 998
00080 FORMAT(' ALPHA COEFFICIENTS')
00090 READ:(A(I),I=1,NN)
00100 PRINT 997
00110 FORMAT(' BETA COEFFICIENTS')
00120 READ:(B(I),I=1,NN)
00130 PRINT 996
00140 FORMAT(' LEAST SQUARE COEFFICIENTS')
00150 READ:(C(I),I=1,NN)
00160 PRINT 1010
00170 FORMAT(' NUMBER, PERIAPSIS VALUES')
00180 READ:N1,(X(I),I=1,N1)
00190 N1=N1-1
00200 PRINT 1015
00210 FORMAT(' PERIAPSIS PROBABILITIES')
00220 READ:(P(I),I=1,N1M1)
00230 PRINT 1020
00240 FORMAT(' NUMBER, VALUES FOR M/CDA')
00250 READ:M,(Z(I),I=1,M)
00260 PRINT 1030

00270 FORMAT(' NUMBER, VALUES FOR TIME IN ORBIT')
00280 READ:N2,(Y(I),I=1,N2)
00290 PRINT 1040
00300 FORMAT(' TIME IN ORBIT PROBABILITIES')
00310 N2=N2-1
00320 READ:(PP(I),I=1,N2M1)
00330 CALL MULPY(N1M1,N2M1,M,JSAVE,HA)
00340 MT=M
00350 MT=MT-1
00360 IF(RP(MT))6,6,7
00370 6 IF(MT-1)7,7,5
00380 7 JM1=MT
00390 T=0.
00400 DO 10 I=1,JM1
00410 T=T+RP(I)
00420 PRINT 1060
00430 FORMAT(' M/CDA DISTRIBUTION')
00440 PRINT (RP(I),I=1,JM1)
00450 PRINT 1070,T
00460 FORMAT('CHECK SUM =',F10.6)
00470 PRINT 1001
00480 FORMAT(IH1)
00490 GO TO 2
00500 END
00510 SUBROUTINE MULPY(N1M1,N2M1,M,JSAVE,HA)
00520 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
00530 COMMON A(20),B(20),C(20),NN,YMEAN
00540 DO 10 I=2,M
00550 RP(I-1)=0.
00560 DO 120 K=1,N1M1
00570 DO 100 I=1,N2M1
00580 T1=TOPH(X(K))
00590 T2=TOPH(X(K+1))
00600 Q1=Y(I)/(HA*T1)
00610 Q2=Y(I+1)/(HA*T1)
00620 Q3=Y(I)/(HA*T2)
00630 Q4=Y(I+1)/(HA*T2)
00640 ZI=MINIF(Q1-Q2,Q3-Q4)
```

REVERSE SIDE


```

00550 T2=TOPH*(K+1)
00560 Q1=Y(I)/(HA*T1)
00570 Q2=Y(I+1)/(HA*T1)
00580 Q3=Y(I)/(HA*T2)
00590 Q4=Y(I+1)/(HA*T2)
00600 ZL=MINIF(Q1,Q2,Q3,Q4)
00610 ZH=MAXIF(Q1,Q2,Q3,Q4)
00620 SP=ZH-ZL
00630 TPR=PP(K)*PP(I)
00640 IF(SP)15,80,20
00650
00660 15 PRINT 1500
00670 1500 FORMAT("INTERVAL NEGATIVE")
00680 60 TO 200
00690 20 DO 30 JL=2,M
00700 IF(Z(JL)-ZL)30,30,25
00710 JTEMP=JL
00720 60 TO 40
00730 30 CONTINUE
00740 40 CONTINUE
00750 DO 60 J=JTEMP,M
00760 IF(Z(J)-ZH)50,45,45
00770 JSAVE=J
00780 ZHIGH=ZH
00790 60 TO 70
00800 50 ZHIGH=Z(J)
00810 RP(J-1)=RP(J-1)+((ZHIG-ZLOW)/SP)*TPR
00820 ZLOW=Z(J)
00830 60 CONTINUE
00840 70 RP(JSAVE-1)=RP(JSAVE-1)+((ZHIG-ZLOW)/SP)*TPR
00850 60 TO 100
00860 80 DO 90 J=2,M
00870 IF(ZL-Z(J))85,82,85
00880 82 RP(J-1)=RP(J-1)+TPR
00890 JSAVE=J
00900 60 TO 100
00910 85 IF(ZL-Z(J))82,90,90
00920 90 CONTINUE
00930 RP(N-1)=RP(N-1)+TPR
00940 JSAVE=M
00950 100 CONTINUE
00960 120 CONTINUE
00970 200 RETURN
00980 END
00990 FUNCTION TOPH(ZZ)
01000 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
01010 COMMON A(20),B(20),C(20),NN,YMEAN
01015 DIMENSION POL(20)
01020 POL(1)=1.
01030 POL(2)=ZZ-A(1)
01040 MN=NN+1
01050 DO 20 I=3,MH
20 POL(I)=ZZ+POL(I-1)-A(I-1)+POL(I-1)-B(I-1)+POL(I-2)
TOPH=YMEAN
DO 25 K=1,NN
25 TOPH=TOPH+C(K)*POL(K+1)
TOPH=EXPF(TOPH)
RETURN
END

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SCIENCE TRADE 2

DELP
1/16/67

```

00000 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
00010 COMMON A(20),B(20),NN,C
00020 PRINT 1000
00030 PRINT 999
00040 FORMAT("1 ORBIT TYPE(1-6)")
00050 FORMAT(" PERIAPSIS DISTRIBUTION PROGRAM")
00060 READ:IT
00065 CALL OTYPE(IT)
00070 A(1)=A(1)/2.
00080 A(NN)=A(NN)/2.
00090 FFM=MM
00100 C=3.14159265/FFM
00110 2 PRINT 1010
00120 FORMAT("1 NUMBER,ANGLE ATTACK VALUES")
00130 READ:N1,(X(I),I=1,N1)
00140 N1=N1-1
00150 PRINT 1015
00160 FORMAT(" ANGLE PROBABILITIES, 90. TO 270.")
00170 READ:(P(I),I=1,N1M1)
00180 PRINT 1020
00190 FORMAT(" NUMBER,VALUES FOR PERIAPSIS DECREMENT")
00200 READ:M,(Z(I),I=1,M)
00210 PRINT 1030
00220 FORMAT(" NUMBER,VALUES FOR VELOCITY INCREMENT")
00230 READ:N2,(Y(I),I=1,N2)
00240 PRINT 1040
00250 FORMAT(" VELOCITY INCREMENT PROBABILITIES")
00260 N2=N2-1
00270 READ:(PP(I),I=1,N2M1)
00280 CALL ADDER(N1M1,N2M1,M,JSAVE)
00290 MT=M
00292 5 MT=MT-1
00294 IF(RP(MT))6,6,7
00296 6 IF(MT-1)7,7,5
00300 7 JM1=MT
00310 T=0.
00320 DO 10 I=1,JM1
00330 T=T+RP(I)
00340 PRINT 1060 PERIAPSIS DECREMENT DISTRIBUTION")
00350 FORMAT("
00360 PRINT:(RP(I),I=1,JM1)
00370 PRINT 1070,T
00380 FORMAT("CHECK SUM =",F10.6)
00390 PRINT 1001
00400 1001 FORMAT(1H1)
00410 GO TO 2
00420 END
00430 SUBROUTINE ADDER(N1M1,N2M1,M,JSAVE)
00440 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
00450 COMMON A(20),B(20),NN,C
00460 DO 10 I=2,M
00470 DO 120 K=1,N1M1
00480 DO 100 I=1,N2M1
00482 T1=PDV(X(K))
00483 T2=PDV(X(K+1))

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FOLDOUT FRAME

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00460 10 RPI-1)=0.
00470 DO 120 K=1,N1M1
00480 DO 100 I=1,N2M1
00482 T1=PDV(X(K))
00483 T2=PDV(X(K+1))
00490 Q:=T1*Y(I)
00500 G2=T1*Y(I+1)
00510 Q3=T2*Y(I)
00520 Q4=T2*Y(I+1)
00530 ZL=MINIF(Q1,Q2,Q3,Q4)
00540 ZH=MAXIF(Q1,Q2,Q3,Q4)
00550 SP=ZH-ZL
00560 TPR=P(K)*PP(I)
00570 IF(SP) 15,80,20
00580 15 PRINT 1500
00590 1500 FORMAT("INTERVAL NEGATIVE")
00600 GO TO 200
00610 DO 30 JL=2,M
00620 IF(Z(JL)-ZL) 30,30,25
00630 25 ZLOW=ZL
00640 JTEMP=JL
00650 GO TO 40
00660 30 CONTINUE
00670 40 CONTINUE
00680 DO 60 J=JTEMP,M
00690 IF(Z(J)-ZH) 50,45,45
00700 45 JSAVE=J
00710 ZHIGH=ZH
00720 GO TO 70
00730 50 ZHIGH=Z(J)
00740 RPI(J-1)=RPI(J-1)+((ZHIGH-ZLOW)/SP)*TPR
00750 ZLOW=Z(J)
00760 60 CONTINUE
00770 70 RPI(JSAVE-1)=RPI(JSAVE-1)+((ZHIGH-ZLOW)/SP)*TPR
00780 GO TO 100
00790 DO 90 J=2,M
00800 IF(ZL-Z(J-1)) 85,82,85
00810 82 RPI(J-1)=RPI(J-1)+TPR
00820 JSAVE=J
00830 GO TO 100
00840 85 IF(ZL-Z(J)) 82,90,90
00850 90 CONTINUE
00860 RPI(M-1)=RPI(M-1)+TPR
00870 JSAVE=M
00880 100 CONTINUE
00890 120 CONTINUE
00900 200 RETURN
00910 END
00920 FUNCTION PDV(ZZ)
00930 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
00940 COMMON A(20),B(20),NN,C
00950 ZZ=(ZZ-90.)/10.
00960 PDV=0.
00970 DO 15 J=1,NN
00980 FJ=J-1
00990 G=C+FJ*ZZ
01000 PDV=PDV+A(J)*COSF(G)+B(J)*SINF(G)
01010 RETURN
01020 END

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FOLDOUT FRAME 2

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00000 C *****PROGRAM TO FIND PROBABILITY OF COMBINATIONS*****
00010 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
00020 PRINT 1000
00030 PRINT 1002
00040 PRINT 1004
00050 FORMAT('1 PROGRAM TO COMBINE RANDOM VARIATES')
00060 FORMAT('0 OPERATION CODE')
00070 FORMAT('0 ADD, 1/" SUBTRACT, 2/" MULTIPLY, 3/"
00080 " DIVIDE, 4/" RESTART, 5 OR GREATER')
00090 FORMAT('0 READ OPERATION CODE(1-4)')
00100 1 PRINT 1010
00110 1010 FORMAT('1 NUMBER,POINTS FOR FIRST DENSITY')
00120 READ:N1,(X(I),I=1,N1)
00130 N1=N1-1
00140 PRINT 1015
00150 FORMAT('FIRST SET OF PROBABILITIES')
00160 READ:(P(I),I=1,N1M1)
00170 PRINT 1020
00180 FORMAT('NUMBER,POINTS FOR RESULTING DENSITY')
00190 READ:M,(Z(I),I=1,M)
00200 2 PRINT 1006
00210 READ:ICODE
00220 IF(ICODE-5)3,1,1
00230 3 PRINT 1030
00240 1030 FORMAT('NUMBER,POINTS FOR NEXT DENSITY')
00250 READ:N2,(Y(I),I=1,N2)
00260 PRINT 1040
00270 1040 FORMAT('NEXT SET OF PROBABILITIES')
00280 N2=N2-1
00290 READ:(PP(I),I=1,N2M1)
00300 CALL ADDER(N1M1,N2M1,M,JSAVE,ICODE)
00310 MT=M
00320 5 MT=MT-1
00330 IF(RP(MT))6,6,7

00340 6 IF(MT-1)7,7,5
00350 7 JM1=MT
00360 T=0.
00370 DO 10 I=1,JM1
00380 T=T+RP(I)
00390 P(I)=RP(I)
00400 PRINT 1060
00410 FORMAT('0 *****RESULTING PROBABILITIES*****')
00420 PRINT 1070,T
00430 1070 FORMAT('CHECK SUM =',F10.6)
00440 N1M1=JM1
00450 DO 20 I=1,M
00460 X(I)=Z(I)
00470 20 X(I)=Z(I)
00480 PRINT 1001
00490 1001 FORMAT('M1)
00500 GO TO 2
00510 END
00520 SUBROUTINE ADDER(N1M1,N2M1,M,JSAVE,ICODE)
00530 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
00540 DO 10 I=2,M
00550 RP(I-1)=0.
00560 DO 120 K=1,N1M1
00570 1 I=1

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SEARCH RESULTS

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005 SUBROUTINE ADDER(N1M1,N2M1,M,JSAVE,ICODE)
00530 COMMON X(100),Y(100),Z(100),P(100),PP(100),RP(100)
00540 DO 10 I=2,M
00550 RP(I-1)=0.
00560 DO 120 K=1,N1M1
00570 DO 100 I=1,N2M1
00580 GO TO (11,12,13,14),ICODE
00590 11 Q1=X(K)+Y(I)
00600 Q2=X(K+1)+Y(I+1)
00610 Q3=X(K)+Y(I+1)
00620 Q4=X(K+1)+Y(I)
00630 GO TO 15
00640 12 Q1=X(K)-Y(I)
00650 Q2=X(K+1)-Y(I+1)
00660 Q3=X(K)-Y(I+1)
00670 Q4=X(K+1)-Y(I)
00680 GO TO 15
00690 13 Q1=X(K)+Y(I)
00700 Q2=X(K+1)+Y(I+1)
00710 Q3=X(K)+Y(I+1)
00720 Q4=X(K+1)+Y(I)
00730 GO TO 15
00740 14 Q1=X(K)/Y(I)
00750 Q2=X(K+1)/Y(I+1)
00760 Q3=X(K)/Y(I+1)
00770 Q4=X(K+1)/Y(I)
00780 15 ZL=MIN1F(Q1,Q2,Q3,Q4)
00790 ZH=MAX1F(Q1,Q2,Q3,Q4)
00800 SP=ZH-ZL
00810 TPR=P(K)+PP(I)
00820 DO 30 JL=2,M
00830 IF(Z(L)-ZL)30,30,25
00840 ZLOW=ZL
00850 JTEMP=JL
00860 GO TO 40
00870 30 CONTINUE
00880 40 CONTINUE
00890 DO 60 J=JTEMP,M
00900 IF(Z(J)-ZH)50,45,45
00910 JSAVE=J
00920 ZHIGH=ZH
00930 GO TO 70
00940 50 ZHIGH=Z(J)
00950 RP(J-1)=RP(J-1)+((ZHIG-H-ZLOW)/SP)*TPR
00960 ZLOW=Z(J)
00970 60 CONTINUE
00980 70 RP(JSAVE-1)=RP(JSAVE-1)+((ZHIG-H-ZLOW)/SP)*TPR
00990 GO TO 100
01000 80 DO 90 J=2,M
01010 IF(ZL-Z(J-1))85,82,85
01020 82 RP(J-1)=RP(J-1)+TPR
01030 JSAVE=J
01040 GO TO 100
01050 85 IF(ZL-Z(J))82,90,90
01060 90 CONTINUE
01070 RP(M-1)=RP(M-1)+TPR
01080 JSAVE=M
01090 100 CONTINUE
01100 120 CONTINUE
01110 200 RETURN
01120 END

```

2

SPHERE
1/26/67

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00000 C*****SPHERICAL PARTICLE PROGRAM*****
00010 COMMON Z(6),X(100)
00020 PRINT 1000
00030 PRINT 1004
00040 1004 FORMAT('0 FUNCTION CODE')
00050 PRINT 1005
00060 1005 FORMAT(' 1, DIAMETER"/' 2, SURFACE AREA')
00070 PRINT 1006
00080 1006 FORMAT(' 3, DRAG PARAMETER"/' 4, VOLUME"/' 5, MASS')
00090 PRINT 1007
00100 1007 FORMAT(' 6, CROSS-SECTIONAL AREA')
00110 1000 FORMAT('1 PROGRAM TO COMPUTE PARTICLE PARAMETERS')
00120 1 CONTINUE
00130 1010 FORMAT(' NUMER, POINTS')
00140 PRINT 1025
00150 1025 FORMAT('1 GIVEN FUNCTION CODE, DENSITY, DRAG')
00160 READ:11,D,C
00170 PRINT 1010
00180 READ:NG,(X(I),I=1,NG)
00190 20 X(I)=Z(I)
00200 PRINT 1040
00210 1040 FORMAT('0 DIAMETER",5X,"SURFACE",7X,"DRAG",6X,"VOLUME"
00220 +,7X,"MASS",7X,"CROSS-SEC")
00230 DO 100 I=1,NG
00240 CALL HORNY(I1,I1,X,I),D,C,Z)
00250 PRINT 1020,(Z(J),J=1,6)
00260 1020 FORMAT(6(1PF12.4))
00270 100 CONTINUE
00280 GO TO 1
00290 END
00300 SURROUTINE HORNY(I1,I1,X,I),D,C,Z)

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000210 10000 FORMAT('00 DIAMETER',5X,'SURFACE',7X,'DRAG',6X,'VOL.11WF'
000220 +,7X,'MASS',7X,'CROSS-SEC')
000230 DO 100 I=1,NC
000240 CALL HORNY(I,11,X(I),D,C,Z)
000250 PRINT 1000,(J),J=1,6)
000260 10000 FORMAT(5(1PF12.4))
000270 100 CONTINUE
000280 GO TO 1
000290 END
000300 SURROUTINE HORNY(INITS,JNITS,Y,D,C,Z)
000310 DIMENSION Z(6)
000320 IX=INITS
000330 JX=JNITS
000340 PI=3.14159265
000350 X=Y
000360 1 GO TO (10,20,30,40,50,60),IX
000370 10 X=PI*X*X
000380 Z(2)=X
000390 11 IX=IX+1
000400 IF(IX-7)18,15,18
000410 15 IX=1
000420 18 IF(IX-JX)1,80,1
000430 20 X=(2.*D*SQR(X/PI))/(3.*C)
000440 Z(3)=X
000450 GO TO 11
000460 30 X=(9.*PI/16.)*(C*X/D)**3
000470 Z(4)=X
000480 GO TO 11
000490 40 X=X*D
000500 Z(5)=X
000510 GO TO 11
000520 50 X=(PI/4.)*(6.*X/(PI*D))*(2./3.)
000530 Z(6)=X
000540 GO TO 11
000550 60 X=SQR(4.*X/PI)
000560 Z(1)=X
000570 GO TO 11
000580 80 CONTINUE
000590 RETURN
000600 END

```

FOURTH PAGE 2