

Federal Aviation Administration

Ground Based Augmentation System Performance Analysis and Activities Report

Reporting Period: July 1 – September 30, 2015

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1. Introduction

The Ground Based Augmentation System (GBAS) team under the direction of the Navigation Branch (ANG-C32) in the Engineering Development Services Division in the Advanced Concepts and Technology Development Office at the Federal Aviation Administration's (FAA) William J Hughes Technical Center (WJHTC) provides this GBAS Performance Analysis / Activities Report (GPAR).

This report identifies the major GBAS related research, testing, and validation activities for the reporting period in order to provide a brief snapshot of the program directives and related technical progress. Currently, the GBAS team is involved in the validation of the GAST-D ICAO SARPs, long-term ionospheric monitoring, supporting system design approval activities for an update to the CAT-I approved Honeywell International (HI) Satellite Landing System (SLS-4000), and maintaining six Ground Based Performance Monitors (GBPMs) and a prototype GAST-D Honeywell Satellite Landing System at Atlantic City International Airport (ACY).

Objectives of this report are:

- a) To provide status updates and performance summary plots per site using the data from our GBPM installations
- b) To present all of the significant activities throughout the GBAS team
- c) To summarize significant GBAS meetings that have taken place this past quarter
- d) To offer background information for GBAS

2. GBAS Updates by Site

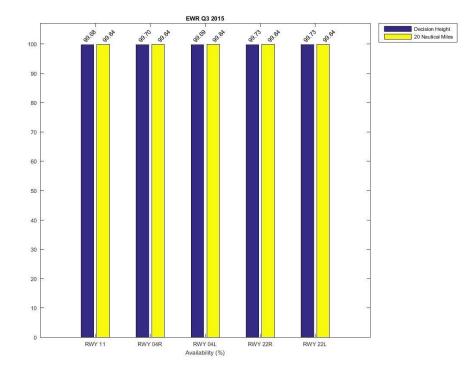
The GBPM was designed and built by ANG-C32 to monitor the performance of our GBAS installations. There are currently six GBPM's in use. They are located in Newark New Jersey (EWR), Houston Texas (IAH), Moses Lake Washington (MWH), Rio de Janeiro Brazil (GIG), and two in Atlantic City New Jersey (ACY). The GBPM is used to monitor integrity, accuracy, availability, and continuity of the LTP and Honeywell's SLS-4000. The plots in each of the following sections utilize a compilation of data collected at one minute intervals. For live, up-to-date data, refer to http://laas.tc.faa.gov. A more detailed description of the GBPM configuration can be found in Appendix D of this report.

2.1 EWR SLS

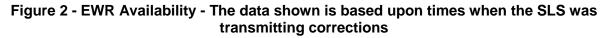
- Newark Liberty Int'l Airport has a Honeywell SLS-4000 that was granted operational approval on September 28, 2012
- Since the EWR SLS-4000 went live until June 2015, there have been 1006 GBAS approaches conducted



Figure 1 - EWR SLS-4000 Configuration



2.1.1 Real Time Performance Data



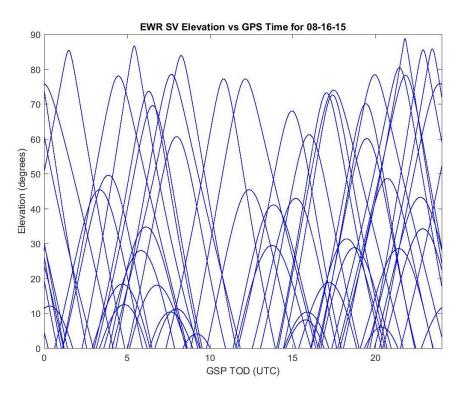
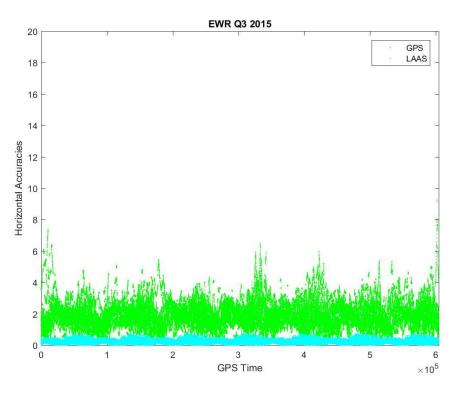


Figure 3 - EWR SV Elevation vs GPS time 08/16/15





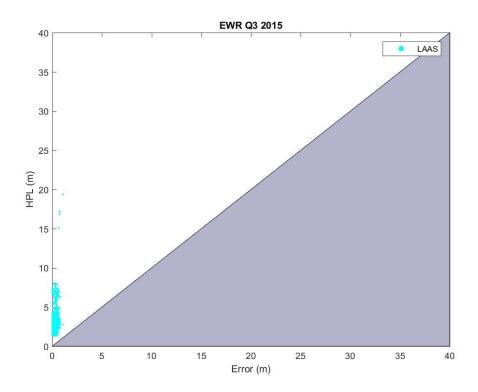
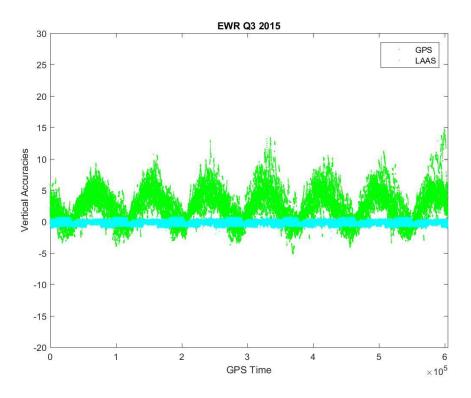
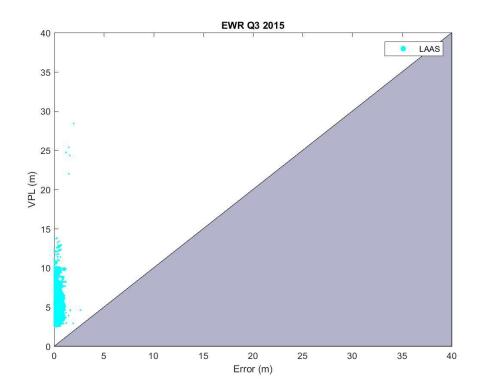


Figure 5 - EWR Horizontal Accuracy vs. Error







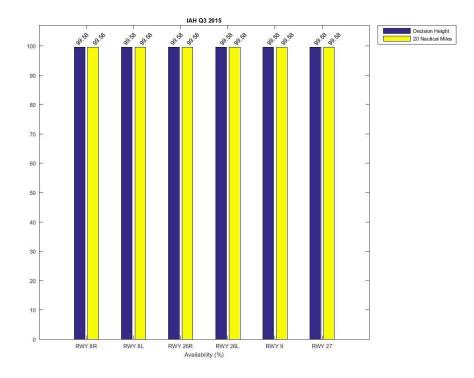


2.2 IAH SLS

- George Bush Intercontinental Airport in Houston, TX has a Honeywell SLS-4000 that was granted operational approval on April 22, 2013
- Since the IAH SLS-4000 went live until June 2015, there have been 1,130 GBAS approaches conducted



Figure 8 - IAH SLS-4000 Configuration



2.2.1 Real Time Performance Data

Figure 9 - IAH Availability - The data shown is based upon times when the SLS was transmitting corrections

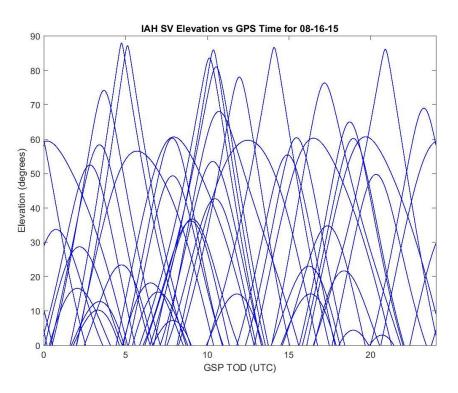
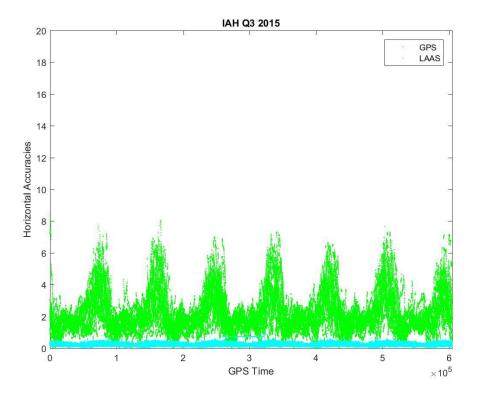
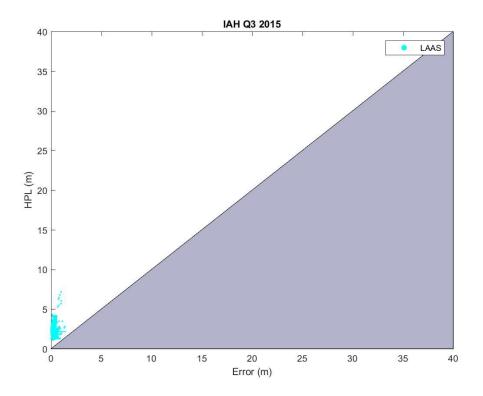


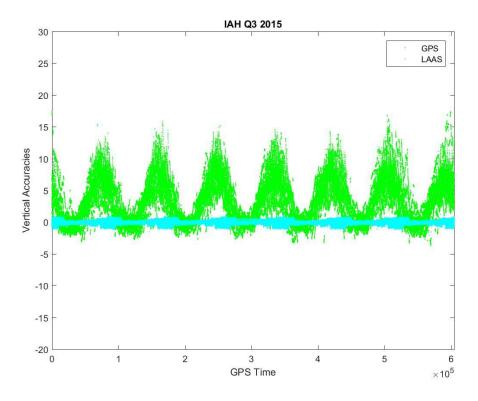
Figure 10 - IAH SV Elevation vs GPS time 08/16/15



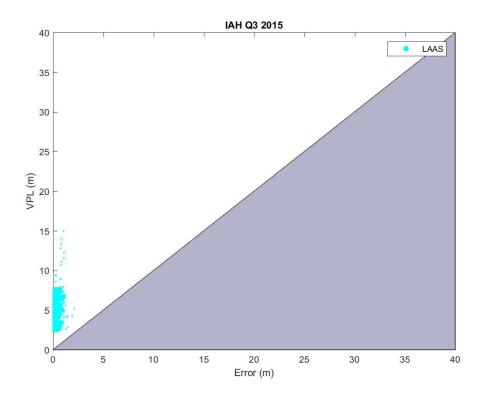












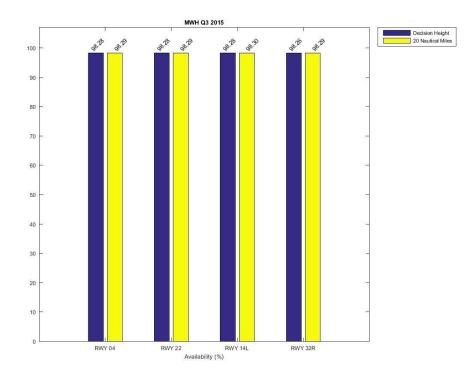


2.3 MWH SLS

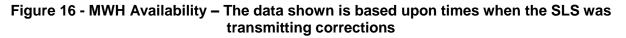
- Grant County Airport in Moses Lake, WA has a private-use Honeywell SLS-4000 that was granted operational approval on January 9, 2013
- Boeing uses this site for aircraft acceptance flights and production activities
- Boeing has also operated this site in a prototype GAST-D mode for flight testing to support GAST-D requirements validation
- While Grant County Airport (GEG) is a public use airport, it has no commercial flights



Figure 15 - MWH SLS-4000 Configuration



2.3.1 Real Time Performance Data



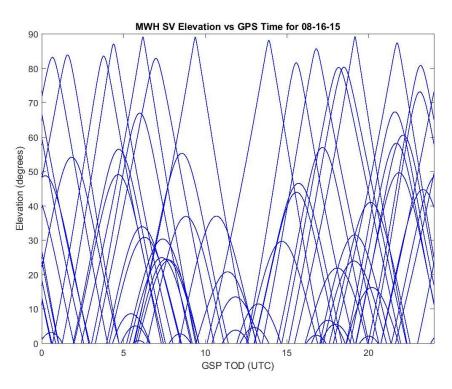


Figure 17 - MWH SV Elevation vs GPS time 08/16/15

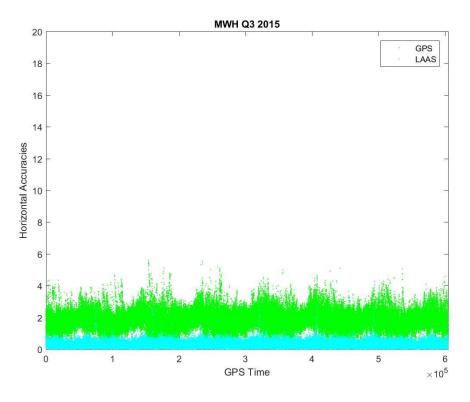


Figure 18 - MWH Horizontal Accuracy Ensemble Plot

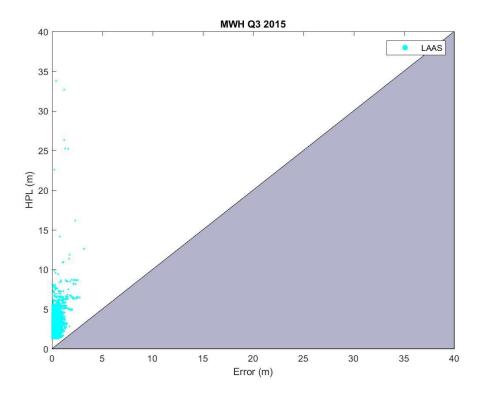
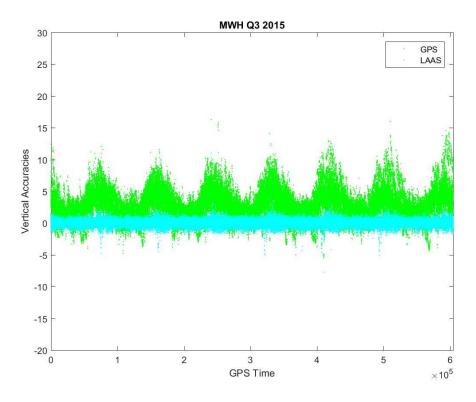


Figure 19 - MWH Horizontal Accuracy vs. Error Bounding Plot





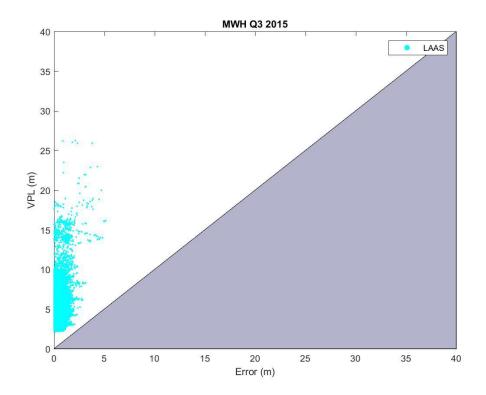


Figure 21 - MWH Vertical Accuracy vs. Error Bounding Plot

2.4 Rio de Janeiro Brazil

- The Rio de Janeiro GBAS is a Honeywell SLS-4000 operating in a CAT-I Block II prototype mode
- The antenna on the Brazil monitor is less robust than the other sites, therefore satellites below 11 degrees may not be tracked as consistently

2.4.1 Real Time Performance Data

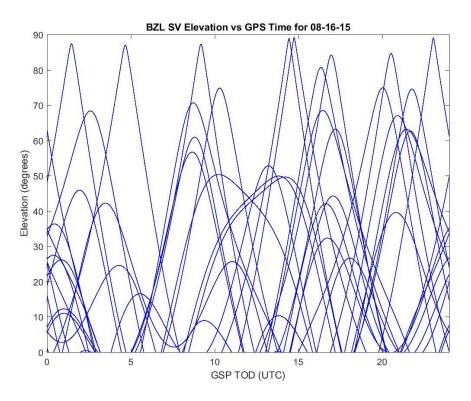


Figure 22 - BZL SV Elevation vs GPS time 08/16/15

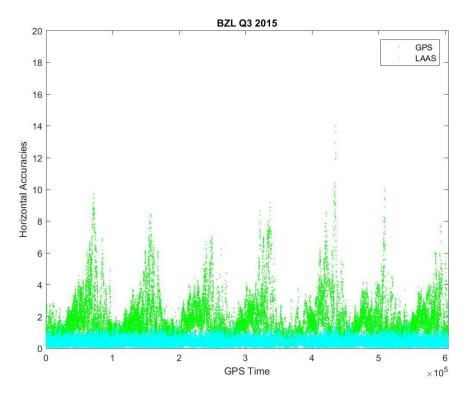


Figure 23 - BZL Horizontal Accuracy Ensemble Plot

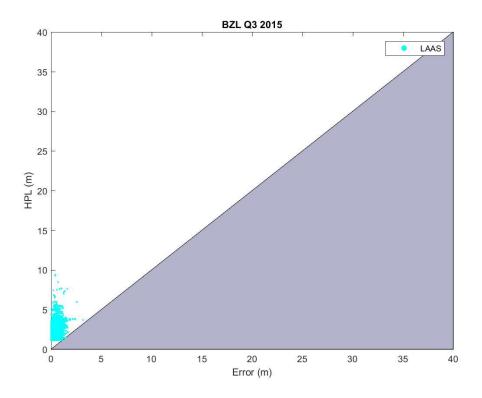
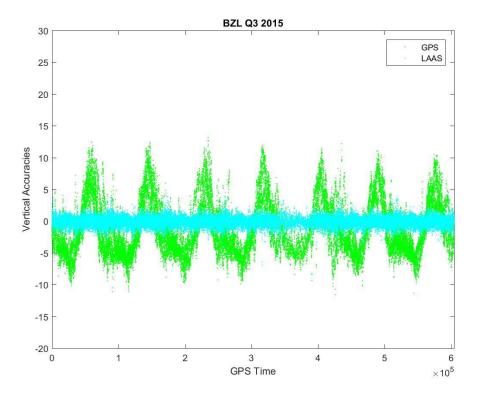


Figure 24 - BZL Horizontal Accuracy vs. Error Bounding Plot





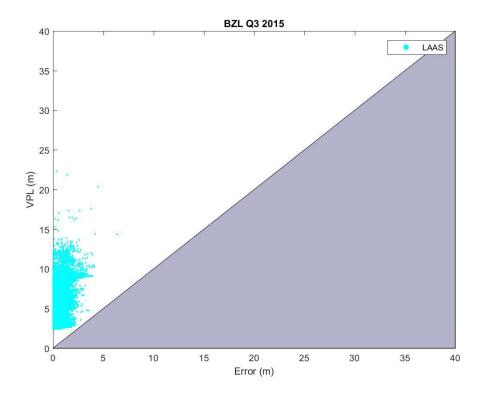


Figure 26 - BZL Vertical Accuracy vs. Error Bounding Plot

2.5 ACY SLS

- The SLS is currently configured for CAT-I Block II for Operational Evaluation and ICMS testing
- See sections 3.2 and 3.2.1 for additional details on the tests performed this quarter

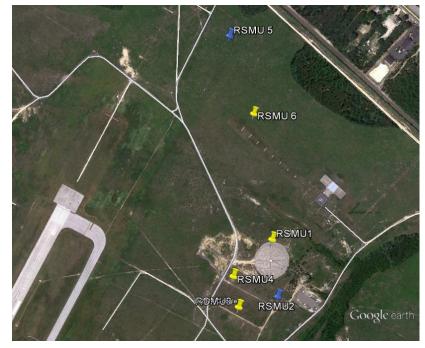
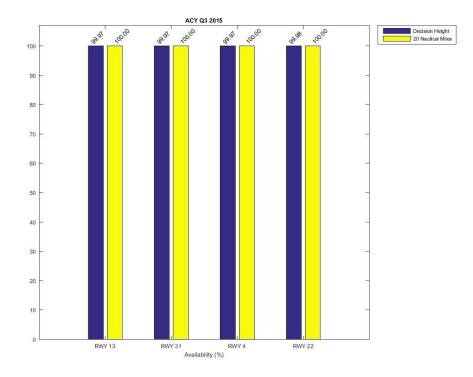
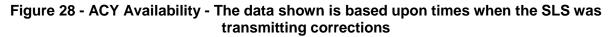


Figure 27 - ACY SLS-4000 Configuration



2.5.1 Real Time Performance Data



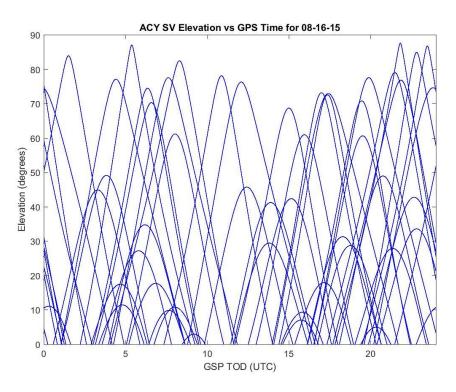


Figure 29 - ACY SV Elevation vs GPS time 08/16/15

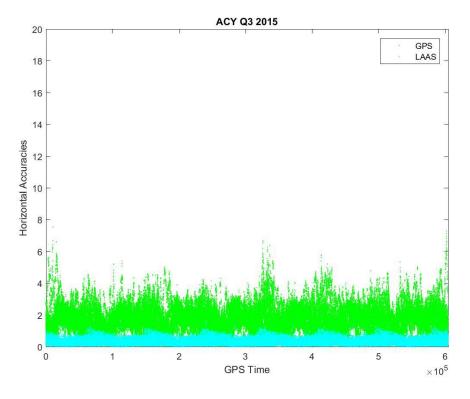


Figure 30 - ACY SLS Horizontal Accuracy Ensemble Plot

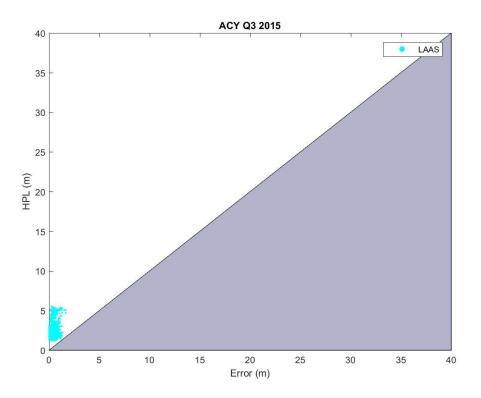
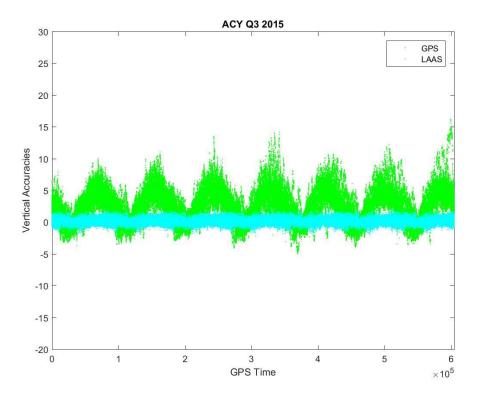


Figure 31 - ACY SLS Horizontal Accuracy vs. Error Bounding Plot





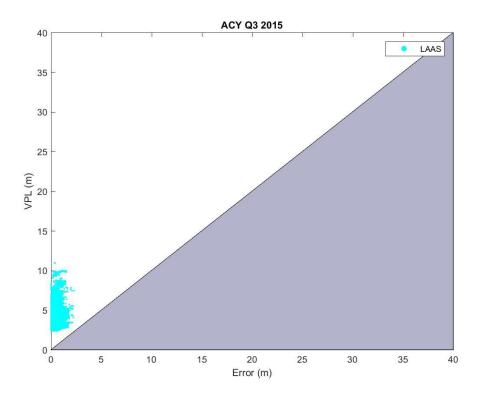


Figure 33 - ACY SLS Vertical Accuracy vs. Error Bounding Plot

2.6 LTP ACY

- The LTP has not been operational this quarter due to ongoing repairs See Appendix C for a full description of the LTP configuration •
- •



Figure 34 - Aerial View of LTP Configuration

3. Research, Development, and Testing Activities

3.1 GBAS GAST-D Validation Status Update

The changes made to the ICAO SARPS to support GAST-D (i.e. GBAS approach service to CAT-III minima using augmentation of the GPS L1 signal) were conditionally accepted at an April 2015 ICAO NSP (Navigation Systems Panel) meeting. More validation work was required for several requirements relating to ionospheric gradient monitoring and VDB compatibility with other systems. Ad-hoc groups have met regularly via teleconference during this quarter to address these issues. ANG-C32 continues to take part in these calls, supporting validation efforts where possible. This includes additional analysis of previously collected ground and airborne data to support open ionospheric gradient monitoring issues, as well as possible testing with existing branch resources to support VDB compatibility work. A final decision on path forward and goal date for closing all open issues is expected in the next quarter.

3.2 Notice Advisory to Navstar Users (NANUs)

The GPS constellation is designed to provide adequate coverage for the continental United States for the majority of the sidereal day. A NANU is a forecasted or reported event of GPS SV outages, and could cause concern if the SV outage(s) creates an insufficient geometry to keep the protection levels below the alert limits. See **Table 4** below for a list of NANU types.

NANUs that caused an interruption in service where Alert Limits are exceeded will be highlighted within the NANU summary (see **Table 5**). Although such an interruption is unlikely, the GBAS team closely tracks the NANUs in the event that post-data processing reveals a rise in key performance parameters.

NANU Acronym	NANU Type	Description
FCSTDV	Forecast Delta-V	Satellite Vehicle is moved during this maintenance
FCSTMX	Forecast Maintenance	Scheduled outage time for Ion Pump Ops / software testing
FCSTEXTD	Forecast Extension	Extends a referenced "Until Further Notice" NANU
FCSTSUMM	Forecast Summary	Gives exact time of referenced NANU
FCSTCANC	Forecast Cancellation	Cancels a referenced NANU
FCSTRESCD	Forecast Rescheduled	Reschedules a referenced NANU
FCSTUUFN	Forecast Unusable Until Further Notice	Scheduled outage of indefinite duration
UNUSUFN	Unusable Until Further Notice	Unusable until further notice
UNUSABLE	Unusable	Closes an UNUSUFN NANU with
		exact outage times
UNUNOREF	Unusable with No Reference NANU	Resolved before UNUSUFN issued
USABINIT	Initially Usable	Set healthy for the first time
LEAPSEC	Leap Second	Impending leap second
GENERAL	General Message	General GPS information
LAUNCH	Launch	Recent GPS Launch
DECOM	Decommission	Removed From constellation

NANU	TYPE	PRN	Start Date	Start Time (UTC)	End Date	End Time (UTC)
2015064	FCSTSUMM	09	07/01/2015	19:51	07/01/2015	22:30
2015065	FCSTMX	04	07/09/2015	03:00	07/09/2015	15:00
2015066	FCSTSUMM	10	07/16/2015	15:00	N/A	N/A
2015067	FCSTSUMM	04	07/09/2015	03:17	07/09/2015	06:13
2015068	LAUNCH	08	07/15/2015	15:36	N/A	N/A
2015069	DECOM	10	07/16/2015	22:28	N/A	N/A
2015070	FCSTDV	17	07/28/2015	16:05	07/29/2015	04:05
2015071	FCSTSUMM	17	07/28/2015	16:24	07/28/2015	21:32
2015072	FCSTDV	19	08/13/2015	03:00	08/13/2015	15:00
2015073	USABINIT	08	08/12/2015	16:53	N/A	N/A
2015074	FCSTSUMM	19	08/13/2015	03:19	08/13/2015	10:55
2015075	FCSTDV	11	08/25/2015	12:20	08/26/2015	00:20
2015076	FCSTSUMM	11	08/25/2015	12:56	08/25/2015	19:37
2015077	FCSTDV	06	09/01/2015	18:40	09/02/2015	06:40
2015078	FCSTSUMM	06	09/02/2015	18:59	09/03/2015	00:51
2015079	FCSTDV	23	09/11/2015	03:25	09/11/2015	15:25
2015080	GENERAL	10	09/16/2015	N/A	N/A	N/A
2015081	FCSTSUMM	23	09/11/2015	03:30	09/11/2015	10:01
2015082	UNUSUFN	14	N/A	N/A	N/A	N/A

Table 1 - NANU Types and Definitions

Table 2 - NANU Summary

4. GBAS Meetings

No meetings reported this quarter.

Appendix A – GBAS Overview

A.1 GBAS Operational Overview

A GBAS is a precision area navigation system with its primary function being a precision landing system. The GBAS provides this capability by augmenting the GPS with real-time broadcast differential corrections.

A GBAS ground station includes four GPS Reference Receivers (RR) / RR antenna (RRA) pairs, and a Very High Frequency (VHF) Data Broadcast (VDB) Transmitter Unit (VTU) feeding an Elliptically Polarized VDB antenna. These sets of equipment are installed on the airport property where a GBAS is intended to provide service. The LGF receives, decodes, and monitors GPS satellite pseudorange information and produces pseudorange correction (PRC) messages. To compute corrections, the ground facility compares each pseudorange measurement to the range measurement based on the survey location of the given RRA.

Once the corrections are computed, integrity checks are performed on the generated correction messages to ensure that the messages will not produce misleading information for the users. This correction message, along with required integrity parameters and approach path information, is then sent to the airborne GBAS user(s) using the VDB from the ground-based transmitter. The integrity checks and broadcast parameters are based on the LGF Specification, FAA-E-3017, and RTCA DO-253D (Airborne LAAS Minimum Operational Performance Standards or MOPS).

Airborne GBAS users receive the broadcast data and use it to compute standardized integrity results. When tuning the GBAS, the user also receives the approach path for navigation with integrity assured. The GBAS receiver applies corrections to GPS measurements and then computes ILS-like deviations relative to the uplinked path providing guidance to the pilot. Airborne integrity checks compare protection levels, computed via the integrity parameters, to alert levels. Protection levels were determined based on allowable error budgets. The horizontal alert limit is 40m and the vertical is 10m at the GAST-C decision height of 200m. If at any time the protection levels exceed the alert limits, calculated deviations are flagged and the approach becomes unavailable. With the current constellation horizontal protection levels are typically 2.3m and vertical protection levels are typically < 5m with resulting availability of 100%.

One key benefit of the GBAS, in contrast to traditional terrestrial navigation and landing systems (e.g., ILS, MLS, TLS), is that a single GBAS system can provide precision guidance to multiple runway ends, and users, simultaneously. Only the local RF environment limits this multiple runway capability. Where RF blockages exist, Auxiliary VDB Units (AVU) and antennas can be added to provide service to the additional runways.

Figure 42 is provided as an illustration of GBAS operation with major subsystems, ranging sources, and aircraft user(s) represented.

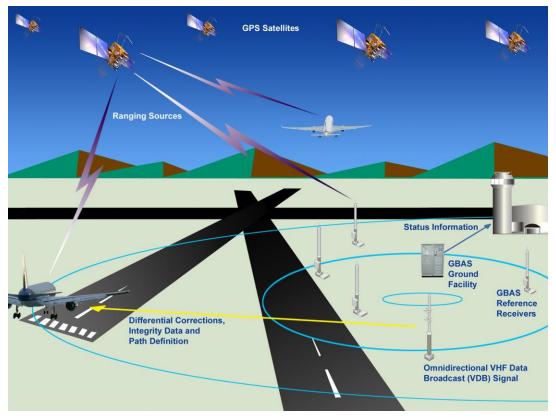


Figure 35 - GBAS Architecture Diagram

Appendix B - GBAS Performance and Performance Type

B.1 Performance Parameters and Related Requirements Overview

The GPS Standard Positioning Service (SPS), while accurate, is subject to error sources that degrade its positioning performance. These error sources include ground bounce multipath, ionospheric delay, and atmospheric (thermal) noise, among others. The SPS is therefore insufficient to provide the required accuracy, integrity, continuity, and availability demands of precision approach and landing navigation. A differential correction, with short baselines to the user(s), is suitable to provide precision guidance.

In addition to accuracy, there are failures of the SPS that are possible, which are not detected in sufficient time and can also cause hazardous misleading information (HMI). GBAS provides monitoring of the SPS signals with sufficient performance levels and time to alarm to prevent HMI.

The relatively short baselines between the user and the GBAS reference stations, as well as the custom hardware and software, is what sets GBAS apart from WAAS. Use of special DGPS quality hardware such as employment of MLA's serves to mitigate the multipath problems, while the GBAS software monitors and corrects for the majority of the remaining errors providing the local user a precision position solution.

The LAAS Ground Facility is required to monitor and transmit data for the calculation of protection parameters to the user. The GBAS specification also requires monitoring to mitigate

Misleading Information (MI) that can be utilized in the position solution. These requirements allow the GBAS to meet the accuracy, integrity, availability, and continuity required for precision approach and landing navigation.

There are three Performance Types (PT) defined within the LAAS Minimum Aviation System Performance Standards (MASPS). The three performance types, also known as Categories, (i.e., Cat I, and Cat II/III), all have the same parameters but with different quantity constraints. For the purposes of this report, the LTP assumes Cat I Alert Limits and hardware classification.

B.2 Performance Parameters

This section highlights the key parameters and related requirements used to depict GBAS system performance in this report. In order to provide the reader a clearer understanding of the plots provided, a little background is being provided below.

Cat I precision approach requirements for GBAS are often expressed in terms of Accuracy, Integrity, Availability, and Continuity. For clarity the use of these four terms, in the context of basic navigation, are briefly described below:

- Accuracy is used to describe the correctness of the user position estimate that is being utilized.
- **Integrity** is the ability of the system to generate a timely warning when system usage should be terminated.
- **Availability** is used to describe the user's ability to access the system with the defined Accuracy and Integrity.
- **Continuity** is used to describe the probability that an approach procedure can be conducted, start to finish, without interruption.

B.2.1 VPL and HPL

Vertical and Horizontal Protection Levels (VPL and HPL) parameters are actively monitored since the GBAS is required to perform with a worst case constellation and geometry scenario. VPL / HPL parameters are directly tied to constellation geometry and when combined with pseudorange errors affect the SPS position estimate and time bias. Monitoring the VPL and HPL in the GBPM gives a valid picture of what the user is experiencing. The protection levels are compared against the alert limits of the appropriate GBAS service level (GSL). In the event the protection levels exceed the alert limit, an outage will occur (See section 6 for GBAS site specific outages).

B.2.2 B-Values

B-values represent the uncorrectable errors found at each reference receiver. They are the difference between broadcasted pseudorange corrections and the corrections obtained excluding the specific reference receiver measurements. B-values indicate errors that are uncorrelated between RRs. Examples of such errors include multipath, receiver noise, and receiver failure.

B.2.5 Performance Analysis Reporting Method

For a given configuration, the LTP's 24-hour data sets repeat performance, with little variation, over finite periods. The GBAS T&E team can make that statement due to the continual processing of raw LTP data and volume of legacy data that has been analyzed from the LTP by the FAA and academia. Constellation and environmental monitoring, in addition to active performance monitoring tools such as the web and lab resources provide the GBAS T&E team indications for closer investigation into the presence, or suspicion, of uncharacteristic performance.

Data sets from the LTP ground and monitoring stations are retrieved on a weekly basis and processed immediately. A representative data-day can then be drawn from the week of data to be formally processed. The resultant performance plots then serve as a snapshot of the LTP's performance for the given week. These weekly plots are afterward compared to adjacent weeks to select a monthly representative set of plots.

Appendix C - LTP Configuration and Performance Monitoring

C.1 Processing Station

The LTP Processing Station is an AOA-installed operational GBAS system. It is continually operational and is used for flight-testing, in addition to data collection and analysis summarized in this report. As an FAA test system, the LTP is utilized in limited modified configurations for various test and evaluation activities. This system is capable of excluding any single non-standard reference station configuration from the corrections broadcast. The performance reporting of the system is represented only from GBAS standard operating configurations.

C.1.1 Processing Station Hardware

The processing station consists of an industrialized Central Processing Unit (CPU) configured with QNX (a UNIX-type real time OS). It then collects raw reference station GPS data messages while processing the data live. It also collects debugging files and special ASCII files utilized to generate the plots found in this report. These collected files are used for component and system level performance and simulation post processing.

The CPU is also configured with a serial card that communicates in real time with the four reference stations through a Lantronix UDS2100 serial-to-Ethernet converter. The reference stations continuously output raw GPS messages to the CPU at a frequency of 2 Hz. Data to and from the reference station fiber lines is run through media converters (fiber to/from copper). The CPU then generates the GBAS corrections and integrity information and outputs them to the VDB.

The VDB Transmitter Unit (VTU) is capable of output of 80 watts and employs a TDMA output structure that allows for the addition of auxiliary VDBs (up to three additional) on the same frequency for coverage to terrestrially or structure blocked areas. The LTP's VTU is tuned to 112.125 MHz and its output is run through a band pass and then through two cascaded tuned can filters. The filtered output is then fed to an elliptically polarized three bay VHF antenna capable of reliably broadcasting correction data the required 23 nautical miles (see Protection Level Maps at http://laas.tc.faa.gov for graphical representation).

Surge and back-up power protection is present on all active processing station components.

C.1.2 Processing Station Software

Ohio University (OU) originally developed the GBAS code through an FAA research grant. Once the code reached a minimum of maturity, OU tested and then furnished the code to the FAA (circa 1996). It was developed using the C programming language under the QNX operating system. QNX was chosen because of its high reliability and real-time processing capability. This LTP code has been maintained by the GBAS T&E team since that time and has undergone numerous updates to incorporate evolving requirements, such as the inclusion of Cat III.

The software stores the precise survey data of the four GBAS reference station antennas (all RRA segments). Raw GPS data (i.e., range and ephemeris info) is received via four GPS receivers. The program cycles through the serial buffers and checks for messages, if one is found, it gets passed to a decoding function. From there, it is parsed out to functions according to message type and the information from the messages is extracted into local LTP variables. Once the system has received sufficient messages, the satellite positions are calculated in relation to the individual reference receivers. Type 1, 2, 4, 11 messages containing differential corrections, integrity values, GS information, and approach path data are then encoded and sent to the VDB via a RS-232 connection. Each of the four message types are encoded separately and sent according to DO-246D standards.

C.2 Reference Stations

There are four reference stations included in the FAA's LTP as required in the GBAS specification. The LTP's reference stations are identified as LAAS Test (LT) sites; there were originally five LT sites (LT1 through LT5), excluding LT4. LT4 was originally used for the L1/L2 site (**Figure 43**).

Each reference station consists of two major component systems. The first is a high quality, GNSS antenna (ARL-1900) manufactured by BAE Systems. The second is the reference receiver.



Figure 36 - The BAE GNSS Multipath Limiting Antenna (MLA)

C.2.1 The BAE ARL-1900 GNSS Multipath Limiting Antenna (MLA)

The BAE Systems ARL-1900 is an innovative, single feed, GNSS antenna that is approximately 6 feet high, and weighs about 35 pounds. The receiving elements are configured in an array, and when combined allow reception of the entire GNSS (Global Navigation Satellite System) band. This antenna is also capable of the high multipath rejection as required by the LAAS specification.

Multipath is a phenomenon common to all Radio Frequency (RF) signals and is of particular concern in relation to DGPS survey and navigation. It is simply a reflection of a primary signal that arrives at a user's equipment at a later time, creating a delay signal that can distort the primary if the reflection is strong. Reflected multipath is the bouncing of the signal on any number of objects including the local water table. Signals that reflect off the earth surface are often referred to as ground-bounce multipath. In all cases, the path length is increased. This path length is critical in GPS since the ranging is based on the signal's Time of Arrival (TOA). This causes a pseudorange error, for the SV being tracked, proportional to the signal strength. The BAE provides at least 23 dB of direct to indirect (up/down) pattern isolation above 5 degrees elevation. These multipath induced pseudorange errors can translate directly into a differential GPS position solution, which would be detrimental to applications such as GBAS. Multipath limiting antennas, such as the BAE Systems ARL-1900, were therefore developed to

address the multipath threat to differential GPS and attenuate the ground multipath reducing the error. The ARL-1900 antenna characteristics also mitigate specular reflections from objects. The antenna's polarization (right hand circular polarized, or RHCP), provides a pattern advantage and reflective LHCP signals, which is left hand circular polarized.

C.3 Multi-Mode Receiver (MMR) Monitoring Station

The GBAS team maintains an MMR on a precise surveyed GPS antenna to monitor ground station performance and evaluate MMR software updates. The MMR drives a dedicated Course Deviation Indicator (CDI). The CDI is a cockpit instrument that indicates fly left/right and up/down information with respect to the intended flight path. A virtual runway was constructed such that the approach path goes through the MMR GPS antenna point. With the configuration, the CDI should always be centered when the MMR is tuned to the virtual runway that coincides with the antenna's survey position. **Figure 44** is a representation of a typical FAA fabricated MMR test/flight user platform. The version of MMR firmware for this reporting period was Flight Change (FC) 31.

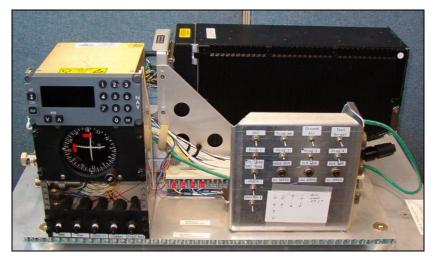


Figure 37 - MMR User Platform

Appendix D - GBPM Configuration and

The Ground Based Performance Monitor is the primary performance monitoring tool for the LTP and the Honeywell SLS-4000 systems. The system uses the received VDB broadcast type 1, 2, 4, and 11 messages from the ground station being monitored along with raw GPS data in order to compute the position of the monitor station. The position calculated from this data is compared to the position of the precision-surveyed GBAS grade GPS antenna, which is used to identify positioning errors.

The GBPM's Novatel OEM-V receiver logs range and ephemeris messages, which provide the necessary pseudorange and carrier phase measurements, as well as satellite position information. VDL messages are then received and separated into each of the DO-246D GBAS message types and decoded.

Data is collected in 24-hour intervals and saved to a .raw file without interruption. This data is used to post-evaluate system performance. In addition to the raw file, live data is transferred from each offsite monitor once per minute to our local database. Users can then access the data through an interactive website by means of tables, charts, and graphs hosted by the Navigation Branch at the FAA. The web address for this service is <u>http://laas.tc.faa.gov</u>.

Analysis of GBPM data is critical for closely observing the LTP and SLS performance behavior. The GBPM data output package contains several plots that can quickly illustrate the overall performance picture of the GBAS. The most useful plots available for performance summary purposes are *Vertical and Horizontal User Error versus Time*. These two plots are often used for preview performance analysis because the "user" GPS sensor position is known and stationary. The known position (precision survey) of the GBPM GPS sensor is compared directly to the computed user position. Typical LTP Vertical and Horizontal user error has an average well within the +/- 1-meter range.

Figure 45 is one of the GBPM's that was built by the Navigation Branch. Some of the major components include a retractable KVM to check the current status of the monitor, CISCO router with a T1 line back to our lab at ACY for data collection and maintenance, Power Distribution Unit (PDU) for a means remote access to bring power outlets back up if they become unresponsive, Novatel GPS Receiver, Becker VDB Receiver, QNX CPU, and an uninterruptable power supply.

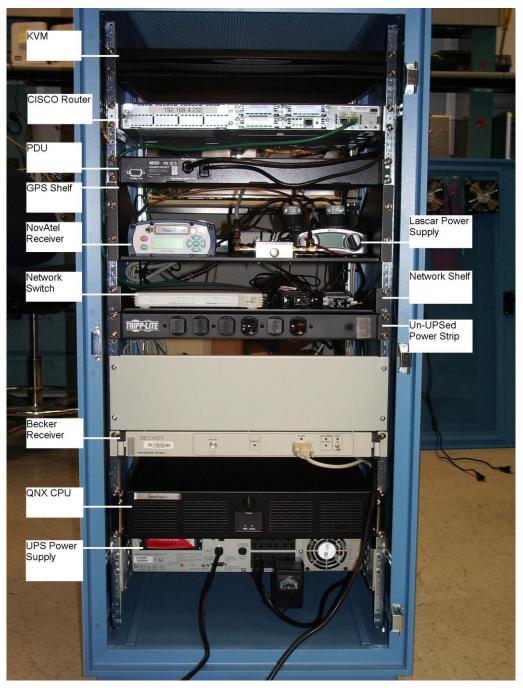


Figure 38 - Ground Based Performance Monitor (GBPM)

Glossary of Terms

ACY
Atlantic City International Airport
—C—
CDI
Course Deviation Indicator
CPU
Central Processing Unit
-E-
—
EWR
Newark Liberty International Airport4
FAA
Federal Aviation Administration
—G—
GBAS
Ground Based Augmentation System 3
GBPM
Ground Based Performance Monitor
GIG
Galeão International Airport4
GNSS
Global Navigation Satellite System
GPAR
GBAS Performance Analysis Report
GSL
GBAS Service Level
-H-
H
Honeywell International
HPL
Horizontal Protection Level
IAH
George Bush Intercontinental Airport
Left Hand Circular Polarized
LT
LAAS Test
—M—
MASPS
Minimum Aviation System Performance Standards
MI

Misleading Information	28
MLA	
Multipath Limiting Antenna	31
MMR	
Multi-Mode Receiver	32
MWH	
Grant County International Airport	4
NANU Nation Advisore to Neuston Lloons	24
Notice Advisory to Navstar Users	24
Ohio University	20
—P—	50
PRC	
Pseudorange Correction	26
PT	20
Performance Type	28
—R—	
RF	
Radio Frequency	31
RHCP	
Right Hand Circular Polarized	32
RRA	
Reference Receiver Antenna	26
—S—	
SLS	
Satellite Landing System	3
SPS	
Standard Positioning Service	27
TOA	2.1
Time Of Arrival	31
—V — VDB	
VDB VHF Data Broadcast	26
VHF Data Broadcast	20
Very High Frequency	26
VPL	20
Vertical Protection Level	28
VTU	20
VDB Transmitter Unit	26
-W-	0
WJHTC	
William J. Hughes Technical Center	3

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