



PHYSICAL AND NUMERICAL SIMULATION OF  
TURBULENT RECIRCULATING FLOWS IN  
MATERIALS PROCESSING OPERATIONS

by

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ABSTRACT

An experimental technique has been developed to make detailed measurements of flow and turbulence characteristics in a heated melt of molten Wood's Alloy using a completely microcomputer controlled data acquisition and processing environment.

A mathematical model has been developed to represent the flow in the melt starting with calculation of the electromagnetic force field and using the  $\kappa - \epsilon$  model of turbulence.

An alternate turbulence model, the XI model, has been developed as part of this work. The new model is computationally simpler and uses fewer semi-empirical constants than the  $\kappa - \epsilon$  model. Comparison of predictions made using the new model with those made using conventional models and also with the experimental measurements is made. The XI model is shown to predict mean flow and turbulence values very close to those predicted by the  $\kappa - \epsilon$  model. Measured values for mean velocities and turbulent dissipation are shown to be reasonably predicted by both the  $\kappa - \epsilon$  and the XI models.

A parameter, PI, is proposed for evaluation of the onset of turbulence in a melt and as an estimate of the degree of turbulence.

A theory, based on fundamental considerations of turbulence, is proposed for explaining the relationship of vessel mixing times with parameters like stirring power input and vessel size. This theory is shown to agree favourably with both experimental and in-plant measurements as quoted in references.

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## 1. INTRODUCTION

In recent years there has been a great increase in the efforts to move towards a reliable description of materials processing operations. This effort is driven, to a large extent, by a desire to automate processes using the now relatively inexpensive computer hardware as "intelligent" controllers. The first step towards computer automation is an accurate mathematical description of the dynamics of the flow and heat and mass transfer characteristics of the system.

This work deals with the physical and numerical modelling of an electromagnetically stirred liquid metal system. The stirring is the result of a force field created by passing a current through the melt between two concentric electrodes, one of which forms the base of the melt while the other dips into the melt from above. This configuration occurs in several processes of industrial importance such as Electro Slag Refining (ESR), Electro Slag Welding (ESW), Submerged Arc Furnace, among others. Many of the problems addressed in this work are, however, more general and consequently many of the results may be applied to a wide range of materials processing operations.

It is felt that the major contributions of this work are:

(1) For the first time, detailed and careful measurements of flow and turbulence parameters have been made in an applied current system using a liquid metal that is solid at room temperature.

(2) A new turbulence model, with an intuitive fundamental basis and a shorter computation time, has been developed and the results compared with existing models and with experiments.

(3) A theory based on fundamental turbulence principles has been developed to explain the dependence of mixing time in metallurgical systems on various operating parameters.

(4) Numerical techniques have been developed to evaluate and correct for the use of thermal anemometers in liquid metals in the presence of magnetic fields and probe surface contamination.

A final objective that this work leads to is the implementation of on-line computer based supervisory process control systems which, it is felt, will become increasingly prevalent in the metals processing industry in the years to come.

Regarding the organisation of this thesis, the material has been divided into six chapters. The Literature Survey is presented in Chapter 2 which reviews work on Thermal

Anemometry, particularly in liquid metals and some experimental and mathematical modelling work on electromagnetically agitated systems and on the theory of turbulence.

Chapter 3 describes the experimental setup and some of the considerations involved in making the measurements.

The formulation and implementation of the numerical models is presented in Chapter 4. Three different models have been used to model the turbulent recirculating fluid flow situation. The first model is a widely used version of the two equation model of turbulence called the k-epsilon model. The second is a low Reynolds number version of the same and the third is a new model developed during this work.

In Chapter 5 the experimental measurements and the numerical predictions are compared and discussed.

The concluding remarks and some suggestions for further work in this and in related areas are made in Chapter 6.

## 2. LITERATURE SURVEY:

In this chapter, literature pertaining to the use of thermal anemometry in liquid metals, to mathematical and physical modelling of electromagnetically driven flows and to turbulent flow phenomena will be reviewed.

### 2.1 Velocimetry

There are many kinds of techniques available for velocity measurement in fluids<sup>(86)</sup>. However, the nature of the experiment imposes several constraints on the types of schemes that may realistically be considered to be possible candidates. The fact that the working fluid is an opaque liquid metal rules out optical techniques like Laser Doppler Anemometry. The fact that relatively high frequency turbulence measurements are required rules out the various forms of Pitot Tube based methods.

There is one instrument whose development and application for measuring turbulent flow have far outstripped those of other instruments up to now, namely, the Thermal Anemometer. The hot wire/film anemometer has been very widely used for measuring turbulent air and water flows and is now a well developed technique for such applications. Its use in measuring liquid metal flows is, however, less well developed. Almost universally, the liquid metal has been liquid mercury

at room temperature, with the exception of Murthy and Szekely<sup>(3)</sup> who used molten Wood's Alloy at 120° C. A large and comprehensive survey of literature pertaining to the use of Thermal Anemometry is presented in (107). This listing contains use of the technique both in liquid metals as well as in more conventional fluids. No effort will be made here to review the use of anemometry in conventional fluids. A sampling of the work pertaining to its use in liquid metal flows is presented in an attempt to point out some of the problems encountered and the solutions adopted by the researchers.

An initial problem regarding the use of thermal anemometry in metals was the fact that the fluid is electrically conducting hence the sensor had to be coated with a film that was both electrically insulating as well as thermally conducting. The pioneering work in developing a practical design of such a probe was done by Sajben<sup>(12,13)</sup> who used a lacquer coated hot wire. In his work Sajben pointed out another serious problem with using hot wire probes in mercury namely one of calibration drift. This problem was acknowledged by most of the other workers in the field<sup>(1,3,7,9,11-13,15,16,18,38)</sup>. Several techniques were proposed by various researchers to overcome this problem.

(1) The drift was observed to occur when the probe crossed the metal-air interface. So the obvious remedy, and one adopted by many of the workers<sup>(1,7,12,13,16,38)</sup>, was to provide for in-situ calibration in the same vessel where the experiment was to be done, so that the probe never had to be lifted out of the metal pool.

(2) In 1969 Hoff<sup>(18)</sup> suggested, based on his experiments, that coating the hot film probe with a sputtered layer of gold or copper improved its wetting characteristics and hence its drift behaviour. However Hurt and Welty<sup>(11)</sup> pointed out that it was difficult to control the uniformity of thickness of the sputter coating and in general concluded that sputter coating was not worth the effort involved because it did not yield any consistent improvement and in fact was found, in some instances, to reduce the life of the probe.

(3) The third approach has involved a heat transfer analysis of the probe, attempting to compensate the fouling of the probe by taking a reading at zero velocity and then subtracting out the effects of the fouling<sup>(1,12,13,3)</sup>.

Another problem associated with liquid metal anemometry was related to the low Prandtl Number of metals. This in fact spawned two, somewhat related, problems. As Hill<sup>(10,14)</sup>



pointed out, the probe in liquid mercury showed poor directional sensitivity except when the Peclet Numbers were very high. This was due to the fact that the thermal boundary layer became more or less uniform around the probe. This observation implied difficulties in the use of cross wire probes to measure flow direction.

A second problem linked to the thick boundary layer was one of reduced frequency response due to inertia effects. Lim et. al.<sup>(6,8)</sup> suggested theoretically calculated high frequency response limits beyond which use of uncompensated outputs from turbulent liquid metal systems may yield erroneous results.

Use of heated liquid metals that are solid at room temperature holds out the possibility of studying solidification problems as linked to fluid flow. Murthy and Szekely<sup>(3)</sup> used modified hot film probes in molten wood's alloy at 125° C. They too encountered the problem of calibration drift which was tackled by using mathematical transformation techniques mentioned earlier.

While most of the workers have concentrated on the problems of anemometry per se in idealised flow situations,

some have employed the technique as a practical tool to investigate flows of metallurgical interest<sup>(1,16)</sup> .

Ricou and Vives<sup>(2)</sup> have employed a completely different technique using an incorporated magnet probe to measure velocities in melts up to 720° C. This technique is applicable in the absence of external magnetic fields and in general cannot be used for turbulence measurements. Also, this very innovative technique is not yet commercially available.

The question of what effect an external magnetic field would have on the performance of a hot film anemometer probe is a very relevant one and has been examined in greater detail in a subsequent section of this chapter.

## 2.2 Mathematical and Physical Modelling

In this section work aimed at modelling certain types of electromagnetically driven flows is reviewed. In 1970 Shercliffe<sup>(26)</sup> studied the case of a point electrode dipping into a semi-infinite pool of inviscid fluid. An analytical solution was developed for this system with a direct current being passed between the point electrode and another electrode set at infinity. Later Souza and Pickering<sup>(27)</sup> extended the analysis for a finite sized current source in a hemispherical

metal pool of finite viscosity. Both these models, though very important from a fundamental standpoint, were rather drastic oversimplifications of the real situation and results could not be extrapolated to metallurgical systems easily. The field was, in a sense, open for the numerical analysts to try and develop a more meaningful description of such systems.

In 1975, Szekely and Asai<sup>(20,21)</sup> established a fully numerical simulation of the turbulent flow and transport in the liquid phase of a continuous caster.

Later, Szekely and Chang<sup>(24,25)</sup> established a numerical calculation for turbulent flow in an induction furnace. Their results for mean velocities were compared with laboratory experiments in a Wood's Alloy system<sup>(23)</sup>. Reasonable agreement (within 30%) was found for mean velocities. Moore and Hunt<sup>(19)</sup> studied turbulent flow in an inductively stirred mercury pool and established measured profiles for turbulence and for mean velocities using a drag probe. Trakas et. al.<sup>(1)</sup> using hot film probes measured spectra in an inductively stirred vessel.

Along somewhat similar lines Szekely and Dilawari<sup>(22,30,32,33)</sup> and Schwerdtfeger and Kreyenberg<sup>(29)</sup> made calculations for temperature profiles, heat transfer and

turbulent velocities for the case of the Electro Slag Refining (ESR) system. This was followed up by Choudhary<sup>(36)</sup> and Choudhary and Szekely<sup>(34,35,39)</sup> who included an automatic calculation of pool profiles in the ESR solidifying ingot on the basis of their heat transfer calculations.

Experimental verification of these calculations has been somewhat less studied. Choudhary<sup>(36)</sup> reported on comparison of measured pool profiles with his calculations. More recently, Choudhary et. al.<sup>(37,31)</sup> reported on comparisons of their calculations with measurements of mean velocity done on the meridional half plane for an applied current mercury system. High speed photography of particle motions on the surface of the mercury yielded somewhat rough measurements of mean velocity. A similar technique was used by Kompan et. al.<sup>(40)</sup> and by Butsenieks et. al. <sup>(43,41)</sup> in a mercury system of related but not identical geometry. Moshnyaga and Sharamkin<sup>(42)</sup> used pitot tube velocimetry to measure the centerline velocities and pressure field in an electromagnetically stirred vessel. Comparison of measurements with a laminar flow numerical simulation was also reported by Butsenieks et. al.<sup>(41)</sup> .

On a considerably more fundamental level, Alemany et. al.<sup>(38)</sup> reported on measurement of turbulent M H D flow

spectra in a behind-grid mercury turbulence with an applied magnetic field.

### 2.3 Turbulence Modelling and Theory

This section relates to the modelling of turbulence phenomena in general. Even though the turbulence model would form an essential component of a mathematical model for the whole process, it has been dealt with separately since many works on turbulence address the idealised problem without much emphasis in the mechanics of the overall flow.

Beek and Miller<sup>(77)</sup> extended some preliminary concepts on the effects of turbulent transport on mixing and reaction rates in chemical reactors. Corrsin<sup>(67,68)</sup> developed expressions relating mixing time for passive scalar contaminants in a turbulent flow to microscales and thence to the stirring power input to the system for the general case of arbitrary Schmidt Number. Brodkey<sup>(69)</sup> applied this analysis to the case of turbulent mixing in a pipe flow and found good agreement between the measured intensity of contaminant segregation and the values predicted using Corrsin's analysis. Brodkey<sup>(70,89)</sup> put forward very physical and intuitively appealing description of the mechanism of contaminant dispersion due to turbulence and some calculations for

parameters like mixing time and segregation based on study of the turbulence spectrum.

Numerical calculation of turbulence can be said to be comprised of the following approaches:

- a) Ad-Hoc viscosity models
- b) One Equation models
- c) Two Equation models
- d) Stress Transport models
- e) Subgrid Scale Closure schemes and Spectral closures

a) and b) are the simplest schemes that are generally considered insufficient to study recirculating flow systems. Two equation models are currently the most popular for recirculating flow situations. These rely on using two parameters to define the state of turbulence in the field. Differential transport equations are written for these two parameters. The choice of parameters for the two equation calculation scheme can yield different models. Of these, the  $\kappa - w$  and the  $\kappa - \epsilon$  models have been the most well known though the latter has recently been the more widely used of the two. Spalding<sup>(65)</sup> first used the  $\kappa - w$  model to predict concentration fluctuations in an axisymmetric gas jet. Rodi<sup>(91)</sup>, Launder and Spalding<sup>(84)</sup> and Rodi and

Spalding<sup>(64)</sup>, among others, described the use of the  $\kappa - \epsilon$  model of turbulence in general two-dimensional flow situations. This model was in fact widely used in most of the numerical work on the simulation of metallurgical flow systems as described in a previous section<sup>(20-25,29-37,39)</sup> and this is among the best documented of all turbulence models.

The stress transport models<sup>(74,75)</sup> use a higher order closure for the turbulence problem resulting in more differential equations to be solved. These models represent the next level of sophistication of turbulence modelling but are presently not very popular because of their greater demands on computer time and general complexity.

Representing the highest level of numerical attack on the turbulence modelling problem are the Spectral Closure schemes, primarily proposed by Orszag and his associates<sup>(63,71,72)</sup> and the Subgrid Closure schemes<sup>(73,76)</sup> which use similar unsteady state schemes but only for flow fluctuations above a certain minimum cutoff eddy size. These methods rely on solving an unsteady form of the Navier-Stokes equations with a very short time step of computation so as to be able to simulate the evolution of the turbulence flow patterns. These techniques are presently used

primarily for meteorological and geophysical flows and represent considerable investment in terms of computer costs.

In 1945, Lotsianskii<sup>(66)</sup> evolved the mathematical concept of a "Disturbance Moment", later known as the Lotsianskii Integral or Invariant, which he showed to be a constant in a homogeneous, isotropic flow and determined by the initial disturbance given to the fluid. Proudman and Reid<sup>(61)</sup> and Batchelor and Proudman<sup>(62)</sup> and Batchelor<sup>(95)</sup> showed that the so called invariant could not truly be a constant but rather a slowly varying function over a flow domain. Comte-Bellot et. al.<sup>(80)</sup> interpreted this property of slow variance to imply a value that was effectively a constant and applied this invariant to study the behaviour of turbulence decaying behind a grid. They showed that this assumption resulted in very reasonable decay laws for grid turbulence. Hinze<sup>(86)</sup> and, to a certain extent, Tennekes and Lumley<sup>(88)</sup> used this together with information about other ranges of a 3-D spectrum to patch together a complete picture of the shape of the isotropic spectrum of turbulence.

Part of this present work has been to use an understanding of spectral shapes, based on the Lotsianskii invariant, together with numerical techniques to evolve a new



scheme of calculation which would fit somewhere between a 1-equation and a 2-equation model of turbulence.

Effects of magnetic fields on turbulence and on a related problem, namely hydrodynamic stability, have been widely studied. Since, in this thesis, the magnetic field strength was judged too weak to have much effect on the turbulence, this part of the literature survey is appended for the sake of completeness. References 44-53, 38 represent some of the work related to this topic.

#### **2.4 Heat and Mass Transfer from a Cylinder in MHD flow**

Thermal anemometer probes when used in the presence of a magnetic field, either aligned with the probe axis or orthogonal to it, experience a change in heat transfer characteristics (ie. a change in the overall Nusselt Number dependence on flow velocity)<sup>(15,38,99)</sup>

Literature surveyed in this section served as a foundation for the numerical calculations that were done in this work to evaluate the influence of a magnetic field on the heat transfer characteristics of a hot wire or hot film probe in a liquid metal flow.

Considerable experimental work has been done in relating the imposition of a magnetic field to the heat transfer from a hot wire/film probe by Lykoudis et. al.<sup>(99)</sup> Numerical and analytical type of solutions were explored since such approaches can be extrapolated to different geometries and sizes easily and also gives a more detailed insight into the mechanics of the problem. Regarding numerical solution of the problem, Kalis et. al.<sup>(96)</sup> considered the effects of a transverse magnetic field on the pressure profile and drag coefficient around a cylinder but did not consider the closely related problem of heat and mass transfer from the cylinder. In addition, as will be shown later, their use of a rather coarse grid near the surface resulted in a markedly weaker dependence of pressure profiles and drag coefficients on the magnetic field.

The analogous heat transfer problem was tackled by Blum<sup>(97)</sup> who used an approximate analytical approach transforming the results obtained for a flat plate boundary layer flow to a circular cylinder.

Chester<sup>(103)</sup> approached analytically the problem of the influence of a magnetic field on flow around a sphere but only in the Stokes' flow regime. Soundalgekar et. al.<sup>(98)</sup> have done numerical calculations for the case of a transverse

magnetic field on a flat plate boundary layer. The general hydrodynamic problem of the numerical analysis of fluid flow around cylinders (albeit in the absence of magnetic fields) has been extensively studied. References 100-102 are a few of the works in this area. Ishiguro et. al.<sup>(104)</sup> and Grosh and Cess<sup>(105)</sup> have reported on experimental measurements of overall heat transfer from a cylinder as related to the flow of liquid metal orthogonal to the cylinder axis. In addition, the books by Hughes and Young<sup>(87)</sup> and by Roache<sup>(85)</sup> contain a wealth of valuable information on the subjects of magneto fluid dynamic modelling and numerical fluid dynamics respectively.

The task to be addressed in this work relates to the measurement and numerical representation of an electromagnetically driven turbulent flow. The literature reviewed here provides a basis for setting up and understanding such a system.

### 3. EXPERIMENTAL MEASUREMENTS

#### 3.1 Introduction

In essence the experiment involved the setting up of a turbulent recirculating flow of liquid metal stirred by passing a direct current through it between two electrodes. The measurements consisted of taking velocity readings at various positions inside the melt and analysing the instantaneous velocity for mean and turbulent components. The anemometer itself, (the principle of operation of which is described in Appendix A) was calibrated using a calibration tow tank. The entire scheme of measurement was completely computer controlled. A brief description of the principal components of this experimental scheme follows. A listing of some of the equipment used in the experimentation, together with the manufacturers' name and location, is given in Appendix D.

#### 3.2 The Main Flow System

The choice of the liquid metal to be used as the principal fluid constituted a major decision and one that had profound effects on the design and selection of equipment. The metal chosen was Wood's Alloy (see Appendix B) Despite the obvious problems associated with operating the vessel and the measurement schemes at higher temperatures, this choice was

made because the use of a metal that was solid at room temperature opened up possibilities of studying solidification linked to fluid flow. This option would clearly not be available if mercury, the generally popular choice, were used. The other advantage was that Wood's Alloy was relatively non-toxic.

The containment vessel consisted principally of a cylindrical copper cylinder (15 cm dia., 10 cm. height ) made from 1/16 inch thick copper sheet and brazed together (see Fig. 3.1). The base electrode was made from 1-1/2 inch thick copper plate that was cut and then turned to final shape. The electrode assembly itself was made in two parts, one being the body of the electrode which covered most of the base of the vessel and was fabricated from the plate as described earlier. The electrical connection was made by a threaded pin which could be screwed into the base plate. This pin provided at the other end a connection point which was turned to the specifications required to match the end of the current carrying cable of the welding generator. The inner walls of the vessel were painted over with Epo-tek H-72 epoxy to prevent current passing through the cylinder walls instead of through the melt entirely. On the outer surfaces of the vessel, a tape heater was fixed using ceramic cement. This tape heater was made in two parts- one for the cylindrical

(vertical) walls of the vessel and the other for the base. The electrical power supply for the two heaters was kept under separate control in order to provide greater flexibility in controlling the melt temperature. The side wall heater was controlled by a proportional temperature controller while the base heater was manually set by potentiostat to create approximately isothermal conditions in the melt. Outside the ceramic jacket which held the electrical heaters was a layer of pyrex cotton wool. This entire assembly was placed inside a Plexiglas outer containment vessel which was held by three support rods above a heavy steel base plate. The three supports could be adjusted to keep the vessel and the metal pool horizontal. Holding the bottom of the vessel several inches above the base plate provided enough space beneath for a drain pot.

For periodic maintenance as well as for any future experiments which may need the vessel to be drained (perhaps to study solidification linked to fluid flow ) the vessel was provided with a drain point. For this purpose, the bottom electrode was made hollow. Through this hole a slender pushrod was passed. A ceramic plug covered the point where the pushrod emerged from the base plate inside the melt to prevent any possible leakage. Draining could be started by tapping the bottom of the pushrod lightly to dislodge the ceramic plug

which would then float to the top of the metal pool. The inner surface of the base plate was made slightly tapered towards the centrally placed drain hole to assist flow of the metal.

The depth of the metal pool was about 4 cms. The top electrode was 1.5 cms. in diameter and dipped into the melt from the Plexiglas upper plate.

Since the flow and turbulence field would be expected to vary spatially inside the melt, a system of accurate probe positioning was required. The positioning equipment was built on top of the upper plate. It consisted of a motor driven radial traverse. On the slider of the radial traverse was attached a probe holder arm which contained a manually operated axial traverse and a simple theta rotation device about the axis of the probe. Both the axial and radial traverse spindles were provided with 10 turn potentiometers which were connected to a digital display. A program was built in the micro-computer to take keyed-in display voltages and interpret them as  $r, x$  coordinates of the probe tip within it. The probe was never moved to a new location until the location was first plotted out and checked for proximity to walls or any solid object which might break the very fragile probe tip. The same program also provided a grid of potentiometer voltage settings. The operator then simply moved the traverses

(radial and axial ) until the approximate voltages appeared on the display.

A sketch of the data flow path for the data relating to the main flow system is given in Fig. 3.2. At the time of the experiment, no attempt was made to analyse or to process the information. The anemometer voltage outputs were continuously recorded on one of the channels of an F.M. tape recorder whose gain factors were set from experience to permit even the highest voltages encountered to be recorded faithfully. The voice memo channel on the recorder was used to make comments about the progress of the experiment and also to note the time of starting and stopping the current and so forth. Time (using a stop watch which was started exactly at the same time the current was switched on) and experiment number was noted on the memo channel at regular intervals to facilitate subsequent playback and analysis. This technique provided a permanent record of the experiment for future analysis and also freed the experimenter from the tasks of operating analysis instrumentation to concentrate on carrying out an experiment which could be potentially dangerous.

The subsequent post-processing of the data took the path as shown in Fig. 3.2. Each tape was played back at least twice, once to get the mean values of voltages for each



experiment and once to do a spectral analysis. The considerations involved in making these two measurements are described in Appendix B. The structure of the micro-computer based data acquisition and information handling scheme is described in Section 3.4.

### 3.3 Calibration System Design

Fig.3.3 shows a sketch of the hot film anemometer calibration assembly, which in essence consisted of a shallow, externally heated, horizontal trough, containing molten Wood's Alloy at 120° C. Provision was made to drag a hot film probe, immersed in the metal, parallel with the axis of the trough at a predetermined velocity. This velocity of traverse ranged from about 1 to 10 cm/sec.

This arrangement is quite similar to that used in the calibration of hot film probes in mercury<sup>(12-14)</sup>, except for the higher operating temperatures involved.

The trough was made from a semi-cylindrical section of 1-1/2 inch copper pipe brazed to two end plates. The cylindrical section of the trough was heated by a semi-cylindrical resistance heater which had an internal diameter equal to the outer diameter of the pipe. The power to

this heater was regulated by a temperature controller. Temperature measurement for the proportional controller was provided by a thermocouple dipping into the pool of metal. A digital display of the temperature was also provided. Movement of the slider assembly was controlled from a remote switching unit and controller. To allow any initial transients to settle down when the probe first started to move, the measurement was started only after the slider had already moved several centimeters. Data acquisition was started (see Fig.3.4) when the slider tripped on a microswitch and stopped when the microswitch tripped off a fixed distance later. This also enabled a velocity measurement as the computer was programmed to measure the 'ON' time of the microswitch. When sampling was over, the velocity as well as the output voltage was printed out by the micro-computer. The rate of sampling could be set in the program. The logic of the data acquisition software and the information handling schemes are detailed in a subsequent section.

### 3.4 Microcomputer Usage Scheme

An Apple II Plus microcomputer was used extensively as a laboratory tool. The use of an in-house microcomputer to supervise the entire experimentation scheme held out several advantages, not the least of which were speed, repeatability,

convenient user interfacing, convenient data handling and also the fact that analog signals needed to be digitised before they could be reduced to any usable form. Functionally, the tasks performed by the microcomputer could be classified as:

- (1) Signal digitisation for both the main system signals and for calibration experiments.
- (2) Preliminary computations like drift correction, scaling etc.
- (3) Information handling involving data file handling and creation, mainframe communications etc.
- (4) Secondary help functions like printing, plotting, interactive human interface, plot digitisation etc.

Many of the program segments performed more than one of the above functions. It was felt that looking at the scheme of microcomputer usage as a step by step sequence performed in a typical experiment would aid in clarifying a somewhat complex chain of interlinked program steps.

Fig. 3.5 shows the Apple II based program files which performed the various operations in a partly sequential manner, while Fig. 3.6 shows the data files that resulted at the end of each segment. A listing of some of the software developed to create this scheme is provided in Appendix E.

These programs are the result of the joint efforts of Mr. T. Kang and this author. The experiments can be divided into three distinct but interlinked data flow paths which are simultaneously processed except at points where data cross linking occurs, in which case a definite priority exists.

#### Path I- Calibration data flow

The microcomputer is connected real time with the calibration arrangement. The anemometer voltage output is led to channel 1 of the Analog-Digital Converter (ADC) and the microswitch is led to channel 5.

##### LEVEL 1(P1L1)

The operating environment is provided by routine DATAQ1 ( in BASIC) and HOT1.OBJ (in Machine Code). HOT1.OBJ was the machine code routine which did the actual data acquisition and stored the new data into specific memory locations. It was called, as a subroutine, by the control routine, DATAQ1 which served as a user interface to input pertinent parameters (like probe location, current etc.) and to set the operating conditions within which HOT1 must operate.

##### LEVEL 2(P1L2)

Routine CALFUL1 (in BASIC) reads the specific memory locations where the data has been stored by the Level 1 routines and

computes velocity of probe traverse and average output voltages during the run. These values are then

- (1) Stored as a disk file under the generic name CALn/m where n represents the probe number and m the calibration number for that particular probe.
- (2) Output to the terminal for viewing
- (3) Output to printer for hard copy

The routine returns control to Level 1 (P1L1) for another run.

#### LEVEL 3 (P1L3)

Routine DRIFT-CALC reads the CALn/m file (once all data points are completed), performs computations to correct the calibration data for drift (described in Section 3.5.3) and stores the corrected values in disk file under the name CALn/m-\*

#### LEVEL 4 (P1L4)

Routine POLYFIT reads the data in data file CALn/m-\* and fits a fourth order polynomial through it. The coefficients are then stored under the generic name of CALn/m-\*-COEF on disk file. This is the final process step in Path-1 and CALn/m-\*-COEF is a data file read by Path2 for processing mean velocities. Thus P1L4 represents an outgoing transfer point.

At this level the calibration curve can be plotted out on the local plotter for viewing, though this is not a necessary step. (See Fig. 5.43)

Path 2- Main Flow System Data Path (Mean Values).

The microcomputer is used in an off-line mode and is used to read data from the taped signals recorded during the experiment.

LEVEL 1 (P2L1)

The environment here is identical to that provided for P1L1. The microswitch is now replaced by a handheld switch for switching on and off the digitisation routine. Operating parameters like rate of sampling or memory allocation may need to be changed according to the needs of the experiment.

LEVEL 2 (P2L2)

Routine SYSFUL1 reads the memory locations for raw data and stores the values of voltages, as well as pertinent data input by the user, in a data file with the generic name SYS-nx where nx stands for the experiment number.

LEVEL 3 (P2L3)

Routine INTERPRET reads the P1L4 data file (CALn/m-\*-COEF) as well as the SYS-nx data file and converts the measured voltages to velocities and also obtains the gradient of the calibration curve at that velocity. This information is stored in file set SYSINT-nx. This stage serves as an outgoing transfer point for Path-2

Path 3- Main Flow System Data Path (Turbulent Values)

## LEVEL 1 (P3L1)

The digitisation and spectral analysis is performed off-line from taped recordings using a HP 5423 A Spectrum Analyser. This level results in an output in the form of a spectral plot on paper.

## LEVEL 2 (P3L2)

Routine SPECTSV is used to digitise the spectral plot. This is done by mounting the plot on the plotter which is then driven along the curve to be traced, by keyboard control. The coordinates of each point on the curve, along with the necessary data for scaling, are saved in file SPEC-nx

## LEVEL 3 (P3L3)

Routine SPECTAN reads data file SPEC-nx and computes values corresponding to turbulence energy and dissipation by numerical integration. These values are saved in file SPECINT-nx.

## LEVEL 4 (P3L4)

Routine COMBINE reads file set SPECINT-nx and also SYSINT-nx (from P2L3) to scale the values of the energies calculated in P3L3 from voltage to velocity based values. The mean velocity (from SYSINT-nx) together with the scaled final values of turbulence energy and dissipation energy along with other pertinent data are stored in data file SYSLAST-nx.

SYSLAST-nx is the final data file which contains only the completely processed experimental results. This file can be either printed locally or transferred to the mainframe computer (IBM-370) using the communication routine T1.

### **3.5 Hot film anemometry in liquid metals**

The use of hot film probes in Wood's Alloy at 120° C posed some definite problems with the sensor itself. Some of the more important problems are described here as well as the steps taken to alleviate them.

The two most important problems were those of probe failure due to the extremely corrosive nature of the heated metal and of the calibration drift with time.

#### **3.5.1 Probe Life**

The temperature of the melt was the single most important reason for failure of the probe. A typical probe as supplied by TSI (without modifications ) would survive about 5-10 minutes in the melt. Damaged probes were cleaned carefully in an ultrasonic bath(80° C, slightly acidic) and then examined using a Scanning Electron Microscope. Over a period of time two failure mechanisms were identified:



(1) The epoxy coating used to insulate the legs of the probe softened at elevated temperatures and quickly eroded away when subjected to a flow of metal.

(ii) The solder connections at the points where the platinum film was joined to the support legs could not withstand the temperature.

After consultations with TSI, the manufacturers of the anemometry equipment, both the epoxy insulation and the solder used at the joints was changed to higher temperature types. The changes have proved extremely satisfactory and probe life extended to at least 40 hours. Even after that time, probes were discarded due to insufficient sensitivity rather than due to failure. A precaution that has been found quite valuable in this regard has been to examine the probe under an optical microscope and touch up the epoxy coating at points where the coating may appear somewhat thin.

### 3.5.2 Calibration Drift

Drift in calibration characteristics was found to occur primarily when the probe was introduced into or removed from the melt. This appears to have been the experience of most other workers in this field while working with mercury. Some

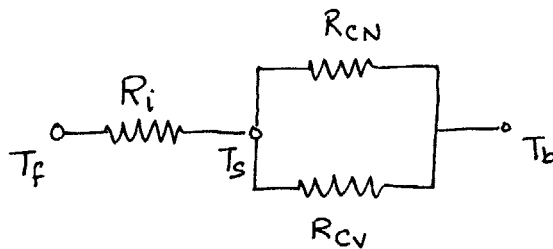
researchers (refer to Chapter 2 for details) have circumvented the problem by crossing the liquid metal -air interface only once during an experiment and by providing arrangements to calibrate the probe in-situ. Practical constraints with the present experimentation forbade this, so a technique was developed to correct for calibration drift using heat transfer analysis. The principal assumption of the analysis was that the drift was directly caused by the appearance of a coating of 'fouling' material (perhaps oxide) on the sensor surface, after Sajben<sup>(12,13)</sup>.

### 3.5.3 Drift Correction Analysis

The drift correction scheme was based on the fact that as the fouling of the sensor surface changed, this change was reflected in the output voltage produced by the anemometer at zero flow velocity ( $E_0$ ; with only natural convection present). Thus the scheme uses the value of  $E_0$ , the zero-flow voltage, as an indicator that points to the extent of fouling. The entire scheme is computed automatically by the microcomputer and results in a new set of calibration points that are generated by resetting the  $E_0$  values for each case to a common base value.

The correction analysis involves three levels of correction, in decreasing magnitude of impact.

Heat is generated in the platinum film by resistance heating and conducted through the quartz coating and fouling material and thence convected away by the fluid. The overall heat transfer stages may be visualised by a simple thermal resistance network.



where:

- $T_f$  is the film temperature
- $T_s$  is the surface temperature
- $T_b$  is the bulk temperature
- $R_i$  is the thermal resistance of fouling material
- $R_{cn}$  is the resistance due to natural convection
- $R_{cv}$  is the resistance due to forced convection
- $Q$  is the total heat flux rate
- $R_p$  is the probe resistance
- $R_s$  is the bridge resistance in series with the probe
- $E$  is the voltage output of the anemometer

$$\Delta T_m = T_f - T_b$$

$$\text{and } \Delta T_{eff} = T_s - T_b$$

$$\Phi = \frac{E^2 R_P}{(R_3 + R_P)^2} = E^2 \beta \quad (3.1)$$

where

$$\beta = \frac{R_P}{(R_3 + R_P)^2} \quad (3.2)$$

Consider first the case when there is no flow. Then:

$$R_{cv} = \infty \quad (3.3)$$

and

$$\Phi_0 = E_0^2 \beta \quad (3.4)$$

where  $E_0$  is the voltage output at zero flow velocity.

Now let:

$$R_{cn} \Delta \eta \Delta T_{eff}^{-1/4} \quad (3.5)$$

$$\Phi_0 = \beta E_0^2 = \frac{\Delta T_{eff}}{R_{cn}} = \frac{\Delta T_{eff}}{\eta \Delta T_{eff}^{-1/4}} \quad (3.6)$$

or

$$\beta E_0^2 = \frac{\Delta T_{eff}^{5/4}}{\eta} \quad (3.7)$$

which gives:

$$\Delta T_{eff}^{5/4} = \eta \beta E_0^2 \quad (3.8)$$

and therefore

$$\Delta T_{eff}^{1/4} = (\eta \beta E_0^2)^{0.2} \quad (3.9)$$

substituting in (3.5) gives

$$R_{cN} = \eta^{.8} \beta^{-.2} E_0^{-.4} \quad (3.10)$$

Now

$$\frac{\Delta T_m}{\Phi_0} = R_i + R_{cN} \quad (3.11)$$

or

$$R_i + \eta^{.8} \beta^{-.2} E_0^{-.4} = \frac{\Delta T_m}{\beta E_0^2} \quad (3.12)$$

introducing the two cases, one the active measurements (superscript (1)) and the other a reference case (superscript (ref)), it may be said that:

$$\Delta R_i = \frac{\Delta T_m}{\beta} \left( \frac{1}{E_0^{(1)}} - \frac{1}{E_0^{(ref)}} \right) - \eta^{.8} \beta^{-.2} \left( E_0^{(1)-.4} - E_0^{(ref)-.4} \right) \quad (3.13)$$

where

$$\Delta R_i \triangleq R_i^{(1)} - R_i^{(ref)} \quad (3.14)$$

Now consider the case when the flow velocity is non-zero. Then  $R_{cv}$  is a finite value.  $R_{cv}$  may be written as

$$R_{cv} = \chi v^{-.42} \quad (3.15)$$

where  $\gamma$  is a constant of proportionality. Here

$$\frac{\Delta T_m}{\beta E_t^{(1)}} = R_i + \frac{f_2 \cdot f_{1,1}}{f_2 + f_{1,1}} \quad (3.16)$$

and

$$\frac{\Delta T_m}{\beta E_t^{(ref)}} = R_i - \Delta R_i + \frac{f_2 \cdot f_{1,ref}}{f_2 + f_{1,ref}} \quad (3.17)$$

where:

$$f_2 \triangleq \gamma V^{-.42} \quad (3.18)$$

$$f_{1,1} \triangleq \gamma \beta^{.8} E_0^{-.2} E_0^{-.4(1)} \quad (3.19)$$

$$f_{1,ref} \triangleq \gamma \beta^{.8} E_0^{-.2} E_0^{-.4(ref)} \quad (3.20)$$

subtracting equation 3.16 from 3.17 yields:

$$\frac{1}{E_t^{(ref)}} - \frac{1}{E_t^{(1)}} = \frac{\beta}{\Delta T_m} \left\{ -\Delta R_i + f_2 \left( \frac{f_{1,ref}}{f_2 + f_{1,ref}} - \frac{f_{1,1}}{f_2 + f_{1,1}} \right) \right\} \quad (3.21)$$

substituting for  $\Delta R_i$  from equation 3.13 gives:

$$\frac{1}{E_t^{(ref)}} - \frac{1}{E_t^{(1)}} = \left( \frac{1}{E_0^{(ref)}} - \frac{1}{E_0^{(1)}} \right) + \frac{\beta}{\Delta T_m} \left\{ f_{1,1} - f_{1,ref} + f_2 \left( \frac{f_{1,ref}}{f_2 + f_{1,ref}} - \frac{f_{1,1}}{f_2 + f_{1,1}} \right) \right\} \quad (3.22)$$

Equation 3.22 is used to calculate values for  $E_t^{(ref)}$ .

The voltages ( $E_0^{(1)}$  and  $E_t^{(1)}$ ) with superscript (1) are the values actually measured during an experiment.  $E_0^{(ref)}$  is the artificial value for base voltage (at zero velocity) selected for normalising the calibration curves. (In

this work,  $E_0^{(ref)}$  has always been kept equal to 15.0 Volts.)  $E_t^{(ref)}$  is the set of normalised voltages which when plotted against the velocity provides a minimum drift calibration curve. This calculation is performed automatically by the computer when a new calibration is done. The resulting (normalised) calibration curve is fitted with a fourth order polynomial. This polynomial is used by other calculation stages in voltage to velocity and turbulent parameter conversions. Figs. 5.42 and 5.43 show a typical case of calibration drift before and after correction was applied.

### 3.6 Effect of Magnetic Fields on Calibration

A question that arose early in the experimental work was that how much does the presence of a magnetic field affect the heat transfer characteristics of the probe. Previous approaches to this problem have included calibrating the probes in the presence of magnetic fields<sup>(38)</sup>, applying semi-empirical correction factors<sup>(15)</sup>, or ignoring the effect of the field altogether. To get a better quantitative appreciation of the problem numerical calculations were made to predict the effect of a transverse magnetic field on the flow of a liquid metal past a heated, infinite cylinder (which represents the hot film probe). The geometry of the flow situation is shown in Fig. 5.36. Consider a circular cylinder

of radius 'a' and of infinite extent in the z-direction. The cylinder surface is heated internally to a constant temperature  $T_f$ . The entire cylinder surface is subjected to a cross flow of a conducting liquid at a velocity  $U_0$  normal to its axis and a constant magnetic field of induction  $B_0$ . The presence of the orthogonal magnetic field alters the velocity profile around the periphery of the cylinder and this in turn influences the heat, mass and momentum transfer characteristics, both local and averaged.

### 3.6.1 Mathematical Formulation

A mathematical formulation of the problem involves the following:

- (1) Momentum balance (Navier Stokes) equations including a source term of electromagnetic origin.
- (2) Maxwell's equations for a moving medium.
- (3) Energy balance equations

The following assumptions were made:

- (1) The fluid is Newtonian
- (2) Property values such as viscosity, density, thermal conductivity etc. were assumed to be constant and evaluated at bulk fluid temperature  $T_{bulk}$ .
- (3) The flow is considered stable and laminar
- (4) The cylinder is an electrical insulator



(5) The cylinder is infinitely long.

The momentum balance equation including the Lorentz force term may be written as:

$$(\underline{U} \cdot \underline{\nabla}) \underline{U} = \frac{\underline{\nabla} p}{\rho} + \nu \nabla^2 \underline{U} + \underline{J} \times \underline{B}$$

and the energy transport equation is given by:

$$(\underline{U} \cdot \underline{\nabla}) T = \alpha \nabla^2 T$$

The effect of the electromagnetic field is introduced into the equation of motion through the  $\underline{j} \times \underline{B}$  term, which as the vector product of magnetic induction and induced current represents the Lorentz force that acts on the fluid. The Lorentz force may be evaluated with the aid of Maxwell's equations, which may be written as:<sup>(87)</sup>

$$\underline{J} = \sigma (\underline{E} + \underline{U} \times \underline{B}) \quad (3.23)$$

$$\underline{\nabla} \times \underline{B} = \mu_0 \underline{J} \quad (3.24)$$

$$\underline{\nabla} \cdot \underline{B} = 0 \quad (3.25)$$

$$\underline{\nabla} \times \underline{E} = - \frac{\partial \underline{B}}{\partial t} \quad (3.26)$$

Expressing the above equations in terms of non-dimensional variables and then rewriting in terms of polar coordinates, yields the following set of equations to be solved:

The momentum balance equation

$$\frac{1}{r} \frac{\partial}{\partial r} \left( \omega \frac{\partial \chi}{\partial \theta} \right) - \frac{1}{r} \frac{\partial}{\partial \theta} \left( \omega \frac{\partial \chi}{\partial r} \right) = \frac{1}{Re} \left[ \frac{\partial}{\partial r} \left( r \frac{\partial \omega}{\partial r} \right) + \frac{1}{r^2} \frac{\partial^2 \omega}{\partial \theta^2} \right] + X_m \quad (3.27)$$

where:

$$X_m = N \left\{ \sin 2\theta \frac{\partial v_r}{\partial r} + \cos 2\theta \frac{\partial v_\theta}{\partial r} \right\} - \omega N \cos^2 \theta \quad (3.28)$$

The vorticity - Stream function relationship:

$$\omega + \frac{1}{r} \frac{\partial}{\partial r} \left( r \frac{\partial \psi}{\partial r} \right) + \frac{1}{r^2} \frac{\partial^2 \psi}{\partial \theta^2} = 0 \quad (3.29)$$

and the energy transport equation:

$$\frac{1}{r} \frac{\partial}{\partial r} \left( T \frac{\partial \psi}{\partial \theta} \right) - \frac{1}{r} \frac{\partial}{\partial \theta} \left( T \frac{\partial \psi}{\partial r} \right) = \frac{1}{Re} \quad (3.30)$$

with the following boundary conditions:

$$\text{at } r = \infty ; \quad 0 \leq \theta \leq \pi \quad \omega = 0 \quad \psi = -r \sin \theta \quad T = T_{\text{bulk}}$$

$$\text{at } \theta = 0, \pi ; \quad 1 \leq r \leq \infty \quad \omega = 0, \quad \psi = 0 \quad \frac{\partial T}{\partial \theta} = 0$$

$$\text{at } r = 1 ; \quad 0 \leq \theta \leq \pi \quad \psi = 0 \quad T = T_w$$

$\omega$  being calculated using first order vorticity generation.

In the above set of equations, the principal variables are defined as follows. Stream function  $\psi$  is defined by:

$$\underline{v} = \nabla \times (\psi \underline{e}_z) \quad (3.31)$$

where

$$\underline{v} = \underline{u} / U_0 \quad (3.32)$$

Vorticity  $\omega$ :

$$\underline{\omega} = \nabla \times \underline{v} \quad (3.33)$$

Interaction parameter  $N$ ;

$$N = \frac{\Gamma B_0^2 a}{\rho U_0} \quad (3.34)$$

Reynold's Number,  $Re$ ;

$$Re = \frac{U_0 a}{\nu} \quad (3.35)$$

Peclet Number,  $Pe$ ;

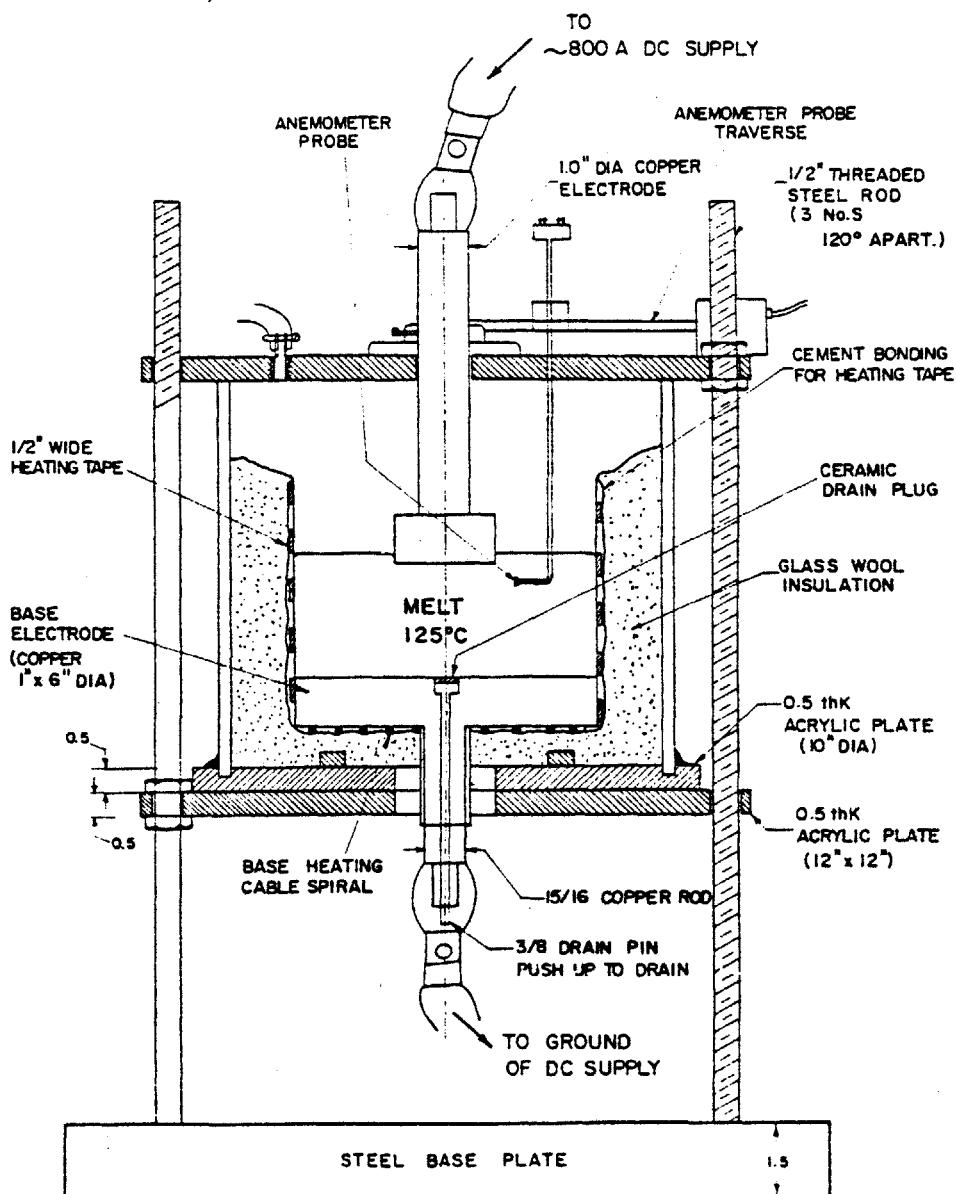
$$Pe = \frac{U_0 a}{\alpha} \quad (3.36)$$

Dimensionless radial distance by

$$r = R/a \quad (3.37)$$

### 3.6.2 Results

The results obtained from the above analysis are discussed in Chapter 5 and in Figs. 5.36 through 5.41.



EXPERIMENTAL SET UP TO STUDY ELECTROMAGNETICALLY DRIVEN LIQUID METAL FLOW

Fig. 3.1 Experimental Set Up for Main Flow System

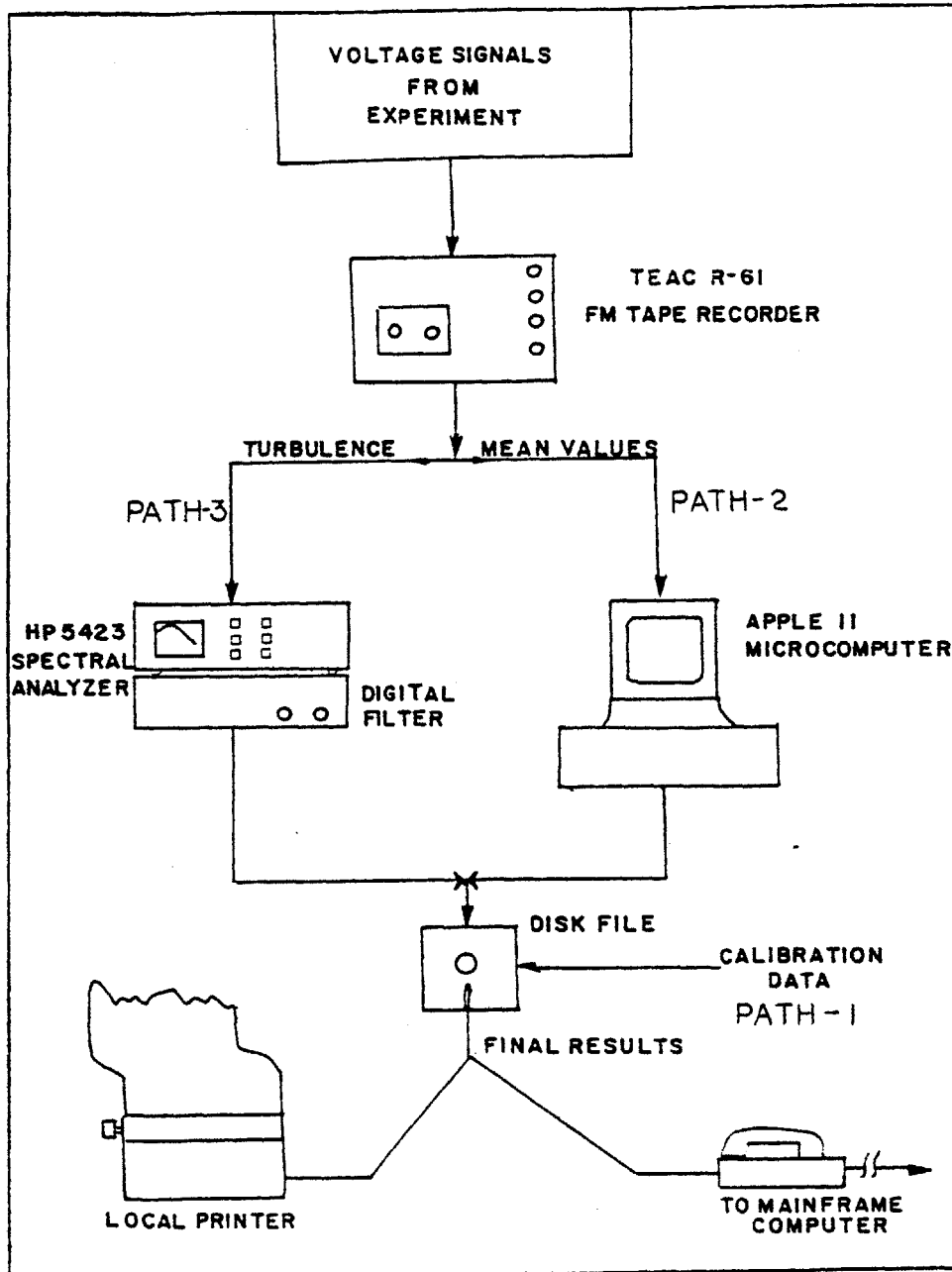


Fig. 3.2 Main Flow System Data Processing Path

## HOT FILM ANEMOMETER CALIBRATION SYSTEM

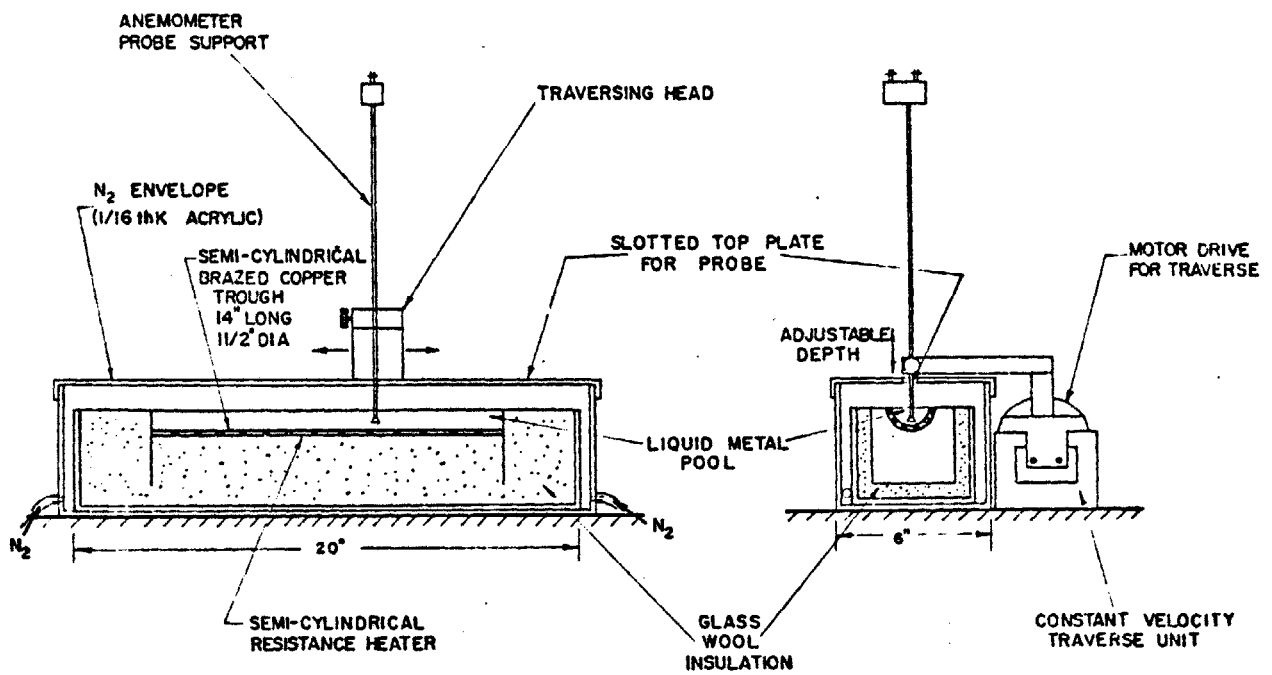


Fig. 3.3 Experimental Set Up for Calibration

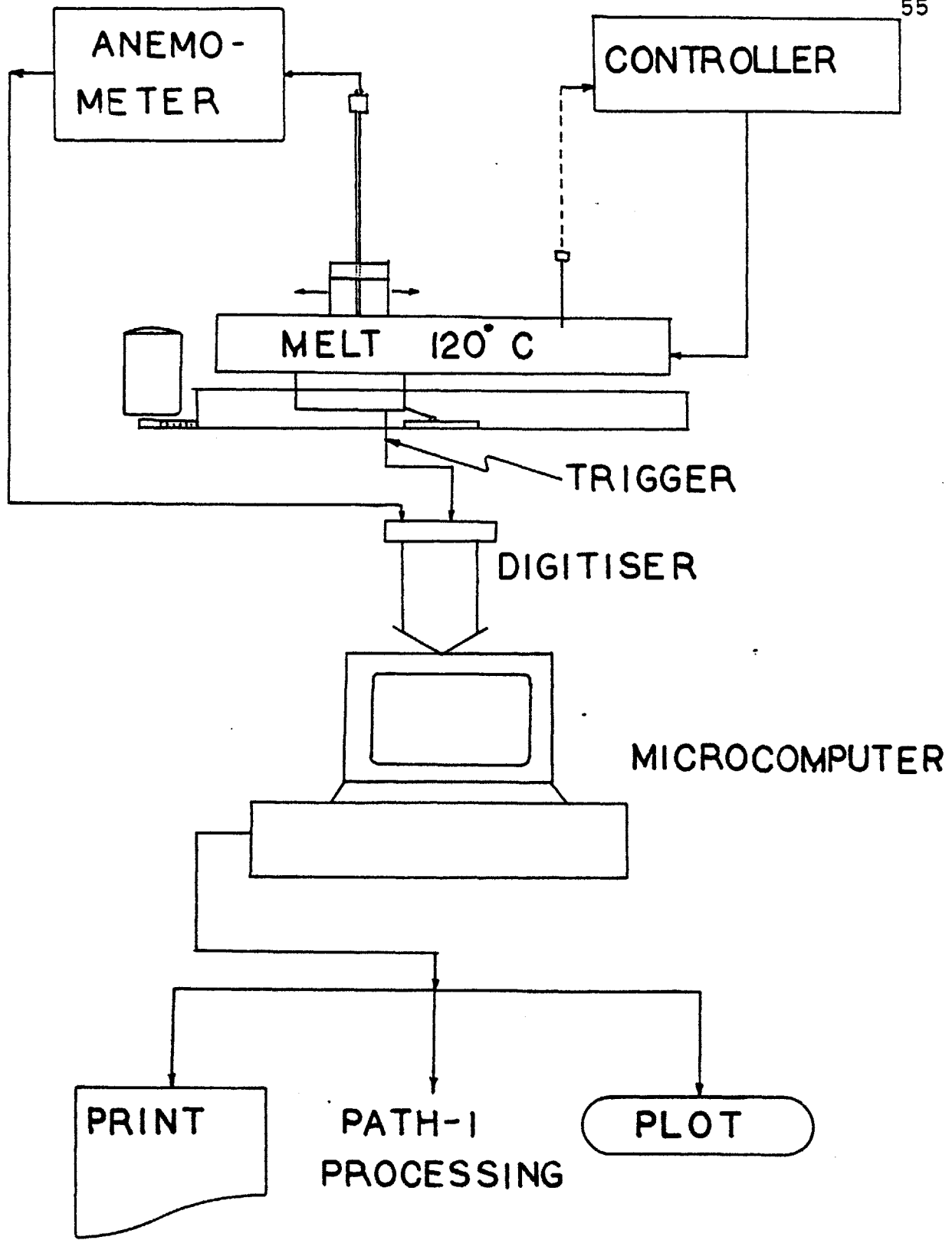


Fig. 3.4 Calibration System Data Processing

Fig. 3.5 Microcomputer Operating Environment - Program Files

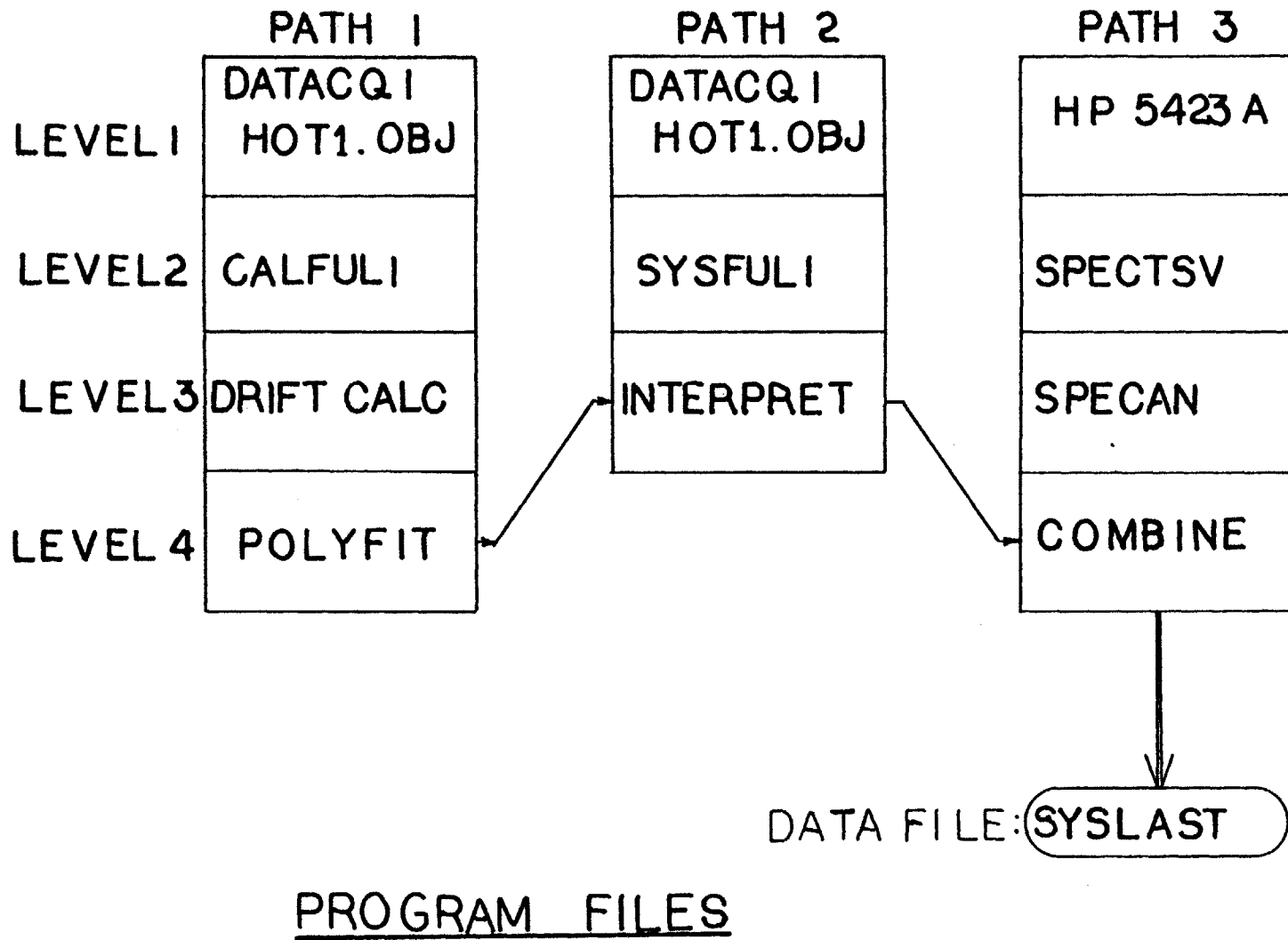
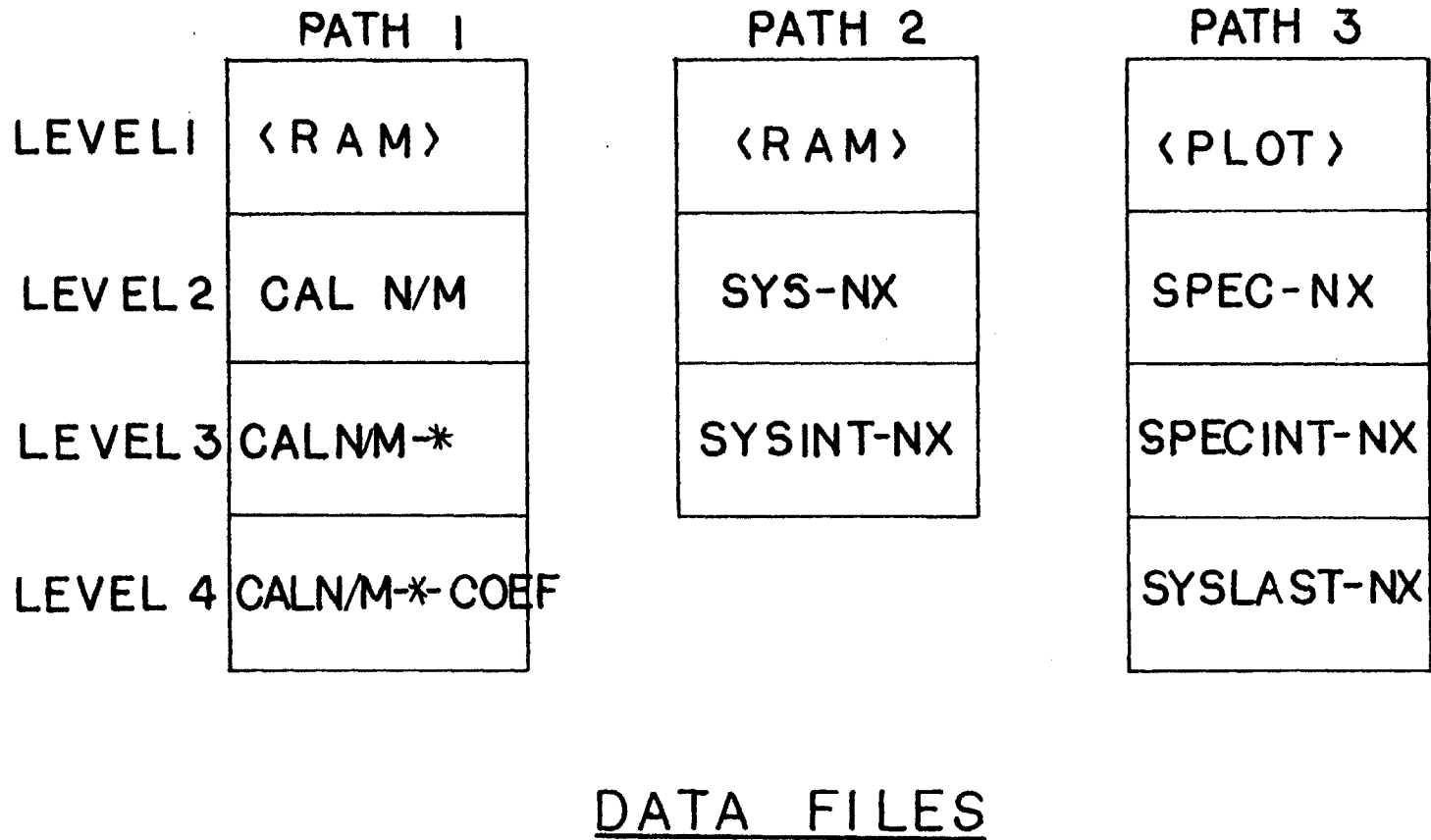




Fig. 3.6 .Microcomputer Operating Environment - Data Files



## 4. THEORETICAL CONSIDERATIONS

### 4.1 Introduction

While results from experimental work possess the advantage of being direct measurements of real physical quantities, they are often restricted to the particular geometry and operating conditions of the experiment. It is therefore difficult, if not impossible, to extrapolate to other operating conditions. A mathematical formulation, coupled to a numerical solution scheme, on the other hand can be extended, almost at will, to differing operating conditions and geometries, within the limits of applicability of the model itself. It also lends itself very directly and simply, to the implementation of microprocessor based process control systems. However, a numerical scheme, needs to be evaluated and checked before acceptance and experimentation provides a convenient vehicle for making such a check.

In this chapter, a general mathematical formulation of the overall problem will first be discussed. Following this, the broad techniques used to convert this formulation to a workable numerical scheme are explained. Modelling of the turbulent flow phenomena has occupied a significant part of the effort involved in this thesis and so will be discussed in some depth. Three different numerical schemes are put forward. The first uses the conventional  $\kappa - \epsilon$  model of turbulence, the

second uses the Jones and Launder<sup>(79)</sup> low Reynolds number model. The third scheme uses a new basis for turbulence computation which was developed as part of this work.

## 4.2 Mathematical formulation of the problem

A mathematical description is sought for the flow phenomena occurring in the Wood's Alloy melt of the geometry and configuration detailed in Chapter 3. A schematic of the flow system is shown in Fig. 4.1

### 4.2.1 Principal Features and Assumptions

The following are the principal features of the system:

- (1) The fluid is Wood's Alloy at 120° C with all the attendant property values which are listed in Appendix B.
- (2) The flow is driven by applying a direct current between unequally sized electrodes. This creates a divergent current path in the melt which in turn creates a recirculating flow in the vessel.
- (3) The flow is recirculating.
- (4) The flow is turbulent.
- (5) The fluid is essentially isothermal.

In formulating the model, the following assumptions were made:

(1) Cylindrical symmetry (no rotational flow) resulting from perfectly centered electrodes and vessel.

(2) The top electrode is assumed to be just touching the surface of the melt. (In the experiment, however, the top electrode was adjusted to be dipping into the melt a few millimetres to prevent any possibility of arcing when the current was turned on.

(3) Quasi steady state.

(4) The flow is assumed to be driven by electromagnetic forces alone, ie. the bouyancy driven component of flow was assumed to be negligibly small.

(5) Electromagnetic damping of the turbulent fluctuations was neglected. Following Alemany et. al.<sup>(38)</sup>, the interaction parameter  $N$ , based on the energy containing eddy size was estimated to be approximately

$$N = \frac{\sigma B_0^2 l}{\rho U} \sim 4 \times 10^{-5} \quad (4.1)$$

which is small enough to have a negligible impact on the spectral energy distribution of the turbulence.

(6) The turbulence was assumed to be isotropic to allow the use of simpler models such as the two-equation model of turbulence.

#### 4.2.2 The governing equations

##### Maxwell's equations

Upon applying the MHD approximation, Maxwell's equations take the following form:<sup>(78)</sup>

Faraday's Law:

$$\underline{\nabla} \times \underline{E} = - \frac{\partial \underline{B}}{\partial t} \quad (4.2)$$

Ampere's Law:

$$\underline{\nabla} \times \underline{H} = \underline{j} \quad (4.3)$$

$$\underline{\nabla} \cdot \underline{B} = 0 \quad (4.4)$$

$$\underline{\nabla} \cdot \underline{j} = 0 \quad (4.5)$$

Furthermore, we have

$$\underline{j} = \sigma (\underline{E} + \underline{v} \times \underline{B}) \quad (4.6)$$

and

$$\underline{B} = \mu_0 \underline{H} \quad (4.7)$$

where  $\sigma$  is the electrical conductivity in mho-m,  $\mu_0$  is the magnetic permeability of free space in Henry/m and  $U$  is the velocity of the fluid in m/s.

Equations 4.2 through 4.5 can be combined<sup>(78)</sup> to give:

$$\frac{\partial \underline{H}}{\partial t} = \eta \nabla^2 \underline{H} + \nabla \times (\underline{U} \times \underline{H}) \quad (4.8)$$

where

$$\eta = \frac{1}{\sigma \mu_0} \quad (4.9)$$

is called the Magnetic Diffusivity. Under conditions of low Magnetic Reynolds' Number,  $R_{em} (= V_0 L \sigma \mu_0)$ , this equation can be reduced<sup>(32,36)</sup> to

$$\frac{\partial \underline{H}}{\partial t} = \eta \nabla^2 \underline{H} \quad (4.10)$$

or

$$\sigma \mu_0 \frac{\partial H_\theta}{\partial t} = \frac{\partial}{\partial r} \left[ \frac{1}{r} \frac{\partial}{\partial r} (r H_\theta) \right] + \frac{\partial^2 H_\theta}{\partial x^2} \quad (4.11)$$

in cylindrical coordinates with axial symmetry with the electromagnetic body force given by:

$$\underline{F}_{be} = \underline{J} \times \underline{B} = \mu_0 \underline{J} \times \underline{H} \quad (4.12)$$

and the current densities being given by:

$$J_r = - \frac{\partial H\theta}{\partial x} \quad (4.13)$$

and

$$J_x = \frac{1}{r} \frac{\partial}{\partial r} (r H\theta) \quad (4.14)$$

### Fluid Flow Equations

Using Reynolds' averaging applied to the turbulent fluctuating velocities, the equation of motion (Navier Stokes' equations) may be written as:

$$\rho(\underline{U} \cdot \nabla) \underline{U} = -\nabla p - \nabla \cdot [\underline{\tau}] + \underline{F}_{be} \quad (4.15)$$

where:

$$\underline{F}_{be} = \underline{1} \times \underline{B} \quad (4.16)$$

and the mass conservation equation:

$$\nabla \cdot \underline{U} = 0 \quad (4.17)$$

Now introducing the vorticity  $\omega$

$$\underline{\omega} = \left[ \frac{\partial U_r}{\partial x} - \frac{\partial U_x}{\partial r} \right] \theta \quad (4.18)$$

and the stream function,  $\psi$

$$U_r = -\frac{1}{r} \frac{\partial \psi}{\partial x} \quad (4.19)$$

and

$$U_x = \frac{1}{\rho r} \frac{\partial \psi}{\partial r} \quad (4.20)$$

the equations 4.15 and 4.17 may be expressed as<sup>(74,36)</sup> :

$$\begin{aligned} r^2 \left[ \frac{\partial}{\partial x} \left( \frac{\omega}{r} \frac{\partial \psi}{\partial r} \right) - \frac{\partial}{\partial r} \left( \frac{\omega}{r} \frac{\partial \psi}{\partial x} \right) \right] - \frac{\partial}{\partial x} \left[ r^3 \frac{\partial}{\partial x} \left( \mu_{eff} \frac{\omega}{r} \right) \right] \\ - \frac{\partial}{\partial r} \left[ r^3 \frac{\partial}{\partial r} \left( \mu_{eff} \frac{\omega}{r} \right) \right] + r^2 \left[ \frac{\partial F_x}{\partial r} - \frac{\partial F_r}{\partial x} \right] \end{aligned} \quad (4.21)$$

where

$$\mu_{eff} = \mu + \mu_t \quad (4.22)$$

The last term can be shown to take the following form:

$$r^2 \left[ \frac{\partial F_x}{\partial r} - \frac{\partial F_r}{\partial x} \right] = - \left[ r \mu_0 H_\theta \Delta r \right] \quad (4.23)$$

In addition the following relationship exists between  $\omega$  and  $\psi$  :

$$\omega + \frac{\partial}{\partial x} \left[ \frac{1}{\rho r} \frac{\partial \psi}{\partial x} \right] + \frac{\partial}{\partial r} \left[ \frac{1}{\rho r} \frac{\partial \psi}{\partial r} \right] = 0 \quad (4.24)$$

The presence of turbulent fluctuations in the flow affects the above formulation through the value for  $\mu_{eff}$ . Considerations for the introduction of turbulent calculations are detailed in section 4.3

### 4.3 Turbulence Modelling - Conventional Models



The presence of turbulence introduces into the stress tensor  $\tau$  (in the time averaged flow equations) an additional term ( $\rho u'v'$ ) called the Reynolds' Stress term. Following Boussinesq<sup>(74)</sup>, turbulent or Reynolds' stresses can be computed using the same relationships which exist for viscous stresses in a Newtonian fluid but by replacing molecular viscosity with a scalar turbulent viscosity. This turbulent viscosity may be calculated using one of the several different methods available. The  $\kappa - \epsilon$  model of turbulence is a two equation model that has been one of the most popular models for calculations of turbulent recirculating flows. In this section the  $\kappa - \epsilon$  model, its low Reynolds' Number version<sup>(72)</sup> and the basis for a new model, the XI model will be examined.

#### 4.3.1 The $\kappa - \epsilon$ Model

The  $\kappa - \epsilon$  model uses two partial differential equations, one for  $\kappa$  and the other for  $\epsilon$ . The  $\kappa$  (turbulence energy) equation can be easily derived by multiplying the (instantaneous) Navier Stokes equations by velocity to form the (instantaneous) Energy equation. Subtracting from this the mean energy equation which can be obtained by multiplying the Reynolds averaged Navier-Stokes equation with the mean

(time averaged) velocity. For high Reynolds Numbers, this

equation reads<sup>(82)</sup>

$$\frac{\overline{1}}{\frac{\partial \kappa}{\partial t}} + \overline{U_i \frac{\partial \kappa}{\partial x_i}} = \frac{\overline{\frac{\partial}{\partial x_i} \left[ U_i \left( \frac{u'_i u'_j}{2} + \frac{P}{\rho} \right) \right]}}{\text{III}} \quad (4.25)$$

where:

$$- \underbrace{\overline{u'_i u'_j} \frac{\partial U_i}{\partial x_j}}_{\text{IV}} - \underbrace{\frac{\epsilon}{\rho}}_{\text{V}}$$

Term I is the time rate of change of  $\kappa$

Term II is the convective transport of  $\kappa$

Term III is the diffusive transport of  $\kappa$

Term IV is the generation of turbulence by mean shear (= 'g')

Term V is the dissipation of turbulence.

Terms II and III are of the redistributive type which just change the local values of  $\kappa$ . Introducing a gradient transport assumption for the diffusive flux of  $\kappa$ , gives the following form of the above equation:

$$U_i \frac{\partial \kappa}{\partial x_i} = \frac{\partial}{\partial x_i} \left( \frac{M_t}{\rho \sigma_\kappa} \frac{\partial \kappa}{\partial x_i} \right) + \frac{M_t}{\rho} \left( \frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) \frac{\partial U_i}{\partial x_j} - \epsilon \quad (4.26)$$

This equation alone cannot be solved because  $\epsilon$  is an unknown variable. This in a sense represents the point of departure for the 1-equation, the 2-equation and the spectral models.

The  $\kappa - \epsilon$  model resolves the problem by postulating another transport partial differential equation for  $\epsilon$  of the form:

$$\frac{\partial \epsilon}{\partial t} + U_i \frac{\partial \epsilon}{\partial x_i} = \frac{\partial}{\partial x_i} \left( \frac{\mu_t}{\rho \sigma_\epsilon} \frac{\partial \epsilon}{\partial x_i} \right) + C_1 \frac{\epsilon}{\kappa} g - C_2 \frac{\epsilon^2}{\kappa} \quad (4.27)$$

where

$\sigma_\epsilon$  is the 'Prandtl Number' for  $\epsilon$  transport

$g$  is the generation of turbulence energy

$C_1$ ,  $C_2$  and  $\sigma_\epsilon$  are constants which were introduced when the exact form of the  $\epsilon$  transport equation<sup>(79)</sup> is reduced to the gradient transport form given above. Constant  $C_2$  is determined from experiments on the decay of behind grid turbulence while constant  $C_1$  is determined from near wall boundary layer flows.

The following are the values suggested<sup>(82,84)</sup> for the constants in the  $\kappa - \epsilon$  model.

$$\begin{aligned} C_1 &= 0.09 & C_2 &= 1.92 & \sigma_\epsilon &= 1.3 \\ C_1 &= 1.44 & \sigma_\kappa &= 1.0 & & \end{aligned}$$

The  $\kappa - \epsilon$  model in the form described above has been applied successfully to many two dimensional wall boundary layers, duct flows, free shear flows and recirculating flows. A useful account of many interesting applications is given by Rodi<sup>(91)</sup> and by Launder and Spalding<sup>(84)</sup>.

#### 4.3.2 The Low Reynolds Number Model

This model due to Jones and Launder<sup>(72)</sup> is an extension of the  $\kappa - \epsilon$  model which was developed primarily to handle near-wall boundary layer flows for computations inside the viscous sublayer. An explanation of the principal basis of this model may be found in reference (72).

#### 4.4 Turbulence Modelling - The XI Model

The XI model is a 1-equation turbulence model which possesses the capability to produce results very similar to, often almost exactly the same as the  $\kappa - \epsilon$  model of turbulence. The technique, in essence, is to eliminate the dissipation energy transport equation altogether and to calculate the local dissipation energy directly from the local kinetic energy at that grid point. This is achieved by utilising certain information about the behaviour of idealised, decaying, isotropic turbulence and extending this to the case of a forced stationary flow. (It is shown later that this is not as strong an assumption as it may seem.) In more physical terms, the technique results in defining a certain quantity (XI) associated with large scale (low wavenumber) eddies which is:

(1) assumed to remain constant over the whole flow field.

(2) determined by the characteristic dimension of the system and by the amount of stirring power supplied to stir the flow system.

In a sense, this implies that the large scale eddies (the so called 'permanent scales') are determined by the dimensions of the flow system to a greater extent than the energy containing scales are, (the latter association being made in the case of conventional 1-equation models.)

#### 4.4.1 Theoretical basis for the model

In 1945, Lotsianskii<sup>(66)</sup> proved that in decaying, isotropic turbulence, a certain functional, the 'Disturbance moment', (later called the 'Lotsianskii Invariant or Integral') should be a constant over the whole flow field and determined by the amount of disturbance supplied to the flow. The functional, defined as

$$I_L = \frac{\pi}{3\pi} \int_0^{\infty} r^4 f(r) dr \quad (4.28)$$

was later shown by Proudman and Reid<sup>(54)</sup> and by Batchelor and

Proudman<sup>(56)</sup> to be not truly a constant but rather a slowly varying function over the flow domain. Comte-Bellot and

Corrsin<sup>(80)</sup> showed that, except under certain conditions of transient flow, the assumption of the constancy of 'I<sub>L</sub>' was reasonable for grid turbulence and indeed produced good results for the decay of behind-grid turbulence.

Using the Lotsianskii integral as determining the low wavenumber end of the spectrum gives a relationship of the form

$$E(k) = I_L k^4 \quad (4.29)$$

for the spectral power density distribution in that region<sup>(77)</sup>

. The spectral shape at higher wavenumber range (in the inertial subrange) has been shown<sup>(77,79)</sup> to be of the form

$$E(k) \propto \epsilon^{2/3} k^{-5/3} \quad (4.30)$$

with the two curves connected by a smooth curve, the Von-Karman patch<sup>(77)</sup> given by

$$E(k) = E(k_e) \frac{(k/k_e)^4}{[1 + (k/k_e)^2]^{17/6}} \quad (4.31)$$

Simple algebraic manipulations of these three expressions yields the location of the energy containing scale,  $k_e$ , as

$$k_e = \left( \frac{1}{I_L} \right)^{3/17} \epsilon^{2/17} \quad (4.32)$$

$$E(k_e) = \frac{1}{2} I_L k_e^4 \quad (4.33)$$

It has also been shown<sup>(77,79)</sup> that integrating the spectral curve

$$\chi = \int_0^{\infty} E(k) dk \quad (4.34)$$

yields an expression of the form

$$E(k_e) = \frac{0.2U'^2}{k_e} \triangleq 0.133 \frac{\chi}{k_e} \quad (4.35)$$

substituting for  $E(k_e)$  from eqn.(4.33) gives

$$\chi = \frac{2}{0.133} I_L k_e^{-17/6} \triangleq I_L k_e^5 \quad (4.36)$$

Substituting for  $k_e$  from eqn.(4.32) gives

$$\chi \triangleq I_L^{2/17} \epsilon^{10/17} \quad (4.37)$$

or

$$\epsilon \triangleq \xi^{1.7} \chi^{1.7} \quad (4.38)$$

where

$$\xi \triangleq (I_L^{-0.118}) \quad (4.39)$$

Equation 4.38 has the capability to yield a point value for dissipation energy, provided that the value of  $\xi$  is known. ( Notice here that the assumption of the Lotsianskii 'disturbance moment'  $I_L$  being a constant is not such a strong assumption since it appears as  $\xi$  which is  $I_L$  raised to the

power of  $-0.118$  or, at worst, as  $\xi^{1.7}$  which is  $I_L$  raised to the power of  $-0.200$ .)

The characteristic length scale associated with the large scale ('permanent') eddies is given by Hinze<sup>(77)</sup> as

$$\left( \frac{I_L}{\mu_t/\rho} \right)^{1/3} \sim L_p \quad (4.40)$$

Now postulating that the largest eddy sizes are constrained by the vessel dimensions and so equating  $L_p$  to the characteristic vessel dimension, ' $L_v$ ', yields an expression of the form

$$\frac{I_L}{\nu_t^2} \triangleq L_v^3 \quad (4.41)$$

$$I_L \triangleq \nu_t^2 L_v^3 \quad (4.42)$$

$$\xi \triangleq \left( \nu_t^2 L_v^3 \right)^{-0.118} \quad (4.43)$$

$\nu_t$ , the characteristic kinematic eddy viscosity can be determined once each iteration cycle by

$$\eta P = \nu_t \rho \iiint_{Vol} [S_{ij}]^2 dVol \quad (4.44)$$

or

$$\nu_t \triangleq \frac{\eta P}{\rho \iiint_{Vol} [S_{ij}]^2 dVol} \quad (4.45)$$

where  $\eta$  is the efficiency of turbulence generation<sup>(60-63)</sup>.

(Also refer to Section 4.6)



Notice that in the process of eliminating the transport equation for  $\epsilon$ , the constants  $C_1$ , and  $C_2$  and  $\sigma_\epsilon$  have also been eliminated. Indeed, in this model, the only constant remaining is  $C_d$  which is left equal to 0.09 as in the  $\kappa - \epsilon$  model (in addition to the Prandtl number for kinetic energy,  $\sigma_\kappa$  which is generally accepted to be approximately unity).

#### 4.5 Boundary Conditions

Boundary conditions for the three different schemes are substantially the same. For the spectral model, however, no boundary conditions for dissipation energy ( $\epsilon$ ) are required since there is no differential equation for this variable.

##### 4.5.1 Boundary conditions for magnetic field calculations

The mathematical statement of the boundary conditions used with the differential equations in the preceding sections is given below. A more detailed explanation of the considerations underlying these statements may be found in references (36) and (78).

(1) Upper Electrode Axial Surface

at  $x=0$ ,  $0 \leq r \leq R_e$

$$\frac{\partial H_\theta}{\partial x} = 0 \quad \left\{ \begin{array}{l} 1_r = 0 \end{array} \right\}$$

(2) Free Surface of Melt

$$\text{at } x=0, R_e \leq r \leq R_m$$

$$H_\theta = \frac{I}{2\pi r} \quad \left\{ \begin{array}{l} J_x = \frac{1}{r} \frac{\partial}{\partial r} (r H_\theta) = 0 \end{array} \right\}$$

(3) Bottom electrode Surface

at  $x=X(5)$ ,  $0 \leq r \leq R_m$

$$\frac{\partial H_\theta}{\partial x} = 0 \quad \left\{ \begin{array}{l} J_r = 0 \end{array} \right\}$$

(4) Cylindrical wall of Vessel

at  $r=R_m$ ,  $0 \leq x \leq X(5)$

$$H_\theta = \frac{I}{2\pi R_m}$$

(5) Centerline of Vessel

at  $r=0$ ,  $0 \leq x \leq X(5)$

$$H_\theta = 0 \quad \text{OR} \quad \frac{\partial}{\partial r} \left( \frac{H_\theta}{r} \right) = 0$$

#### 4.5.2 Boundary Conditions for Flow Equations

Boundary conditions are required for mean velocities and also for turbulence parameters. The conditions for the mean values should be stated in terms of the primary variables, Vorticity ( $\omega$ ) and Stream Function ( $\psi$ ). The conditions for turbulence parameters are given in terms of expressions for kinetic energy ( $\kappa$ ) and (except in the case of the spectral model) dissipation energy ( $\epsilon$ ). Only the final form of the

boundary conditions are given here. A more detailed explanation may be found in references (36,74-76)

(1) Centerline of Vessel

at  $r=0$ ,  $X(1) \leq x \leq X(5)$

$$\psi = \frac{\partial \pi}{\partial r} = \frac{\partial \epsilon}{\partial r} = 0$$

$$\left(\frac{\omega}{r}\right)_0 = \frac{g}{\rho} \left[ \frac{(\gamma_0 - \gamma_2)}{r_2^2} + \frac{(\gamma_1 - \gamma_0)}{r_1^2} \right] / (r_2^2 - r_1^2)$$

Where suffixes 0,1,2 denote the points on the centerline axis and the adjacent grid nodes in the r-direction respectively.

(2) Free Surface of Melt

at  $x=X(1)$ ,  $R_e \leq r \leq R_m$

$$\psi = \frac{\omega}{r} = \frac{\partial \pi}{\partial x} = \frac{\partial \epsilon}{\partial x} = 0$$

(3) Top Electrode Axial Surface

at  $x=X(1)$ ,  $0 \leq r \leq R_e$

$$\psi = 0, \quad \pi = \epsilon = 0$$

$$\left(\frac{\omega}{r}\right)_0 = \frac{3(\gamma_0 - \gamma_1)}{\rho r^2 (x_1 - x_0)^2} - \frac{1}{2} \left(\frac{\omega}{r}\right)_{r_1}$$

Where suffixes 0 and 1 refer to a grid node on the boundary and to the adjacent node in the x-direction respectively.

(4) Bottom Electrode Axial Surface

at  $x=X(5)$ ,  $0 \leq r \leq R_m$

$$\psi = 0, \quad \pi = \epsilon = 0$$

$$\left(\frac{\omega}{r}\right)_0 = \frac{3(\gamma_0 - \gamma_1)}{\rho r^2 (x_1 - x_0)^2} - \frac{1}{2} \left(\frac{\omega}{r}\right)_{r_1}$$

The suffixes have the same meaning as in B.C. 4 above

(5) Cylindrical Wall

at  $r=R_m$ ,  $X(1) \leq x \leq X(5)$

$$\pi = \epsilon = 0$$

$$\left(\frac{\omega}{r}\right)_0 = \frac{3}{P} \frac{(Y_0 - Y_1)}{(r_1 - r_0)^2 r_0 r_1} - \frac{1}{2} \left(\frac{\omega}{r}\right)_1$$

Where suffixes 0 and 1 refer to a grid point on the boundary and to the adjacent node in the r-direction respectively.

#### 4.6 Numerical Solution Scheme

The derivation of the finite difference discretization equations and the treatment of wall and neighbor nodes follows the techniques laid down in references (74) and (36) and will not be discussed here. The program itself is based, in part, on the work of Dr. M. K. Choudhary<sup>(36)</sup>.

Treatment of boundary conditions for the turbulent parameters needs special care. As detailed by Pun and Spalding<sup>(82)</sup>, a pseudo boundary condition is set up for the wall neighbor nodes. A source term for the kinetic energy equation is specified as:

$$S_k = \tau_w \frac{\partial u}{\partial n} - C_d \frac{\rho^2 \pi^2}{\tau_w} \frac{\partial u}{\partial n} \quad (4.46)$$

at the neighbor node while the dissipation energy is calculated from :

$$\epsilon = C_d^{3/4} \pi^{3/2} / (0.41 \delta) \quad (4.47)$$

The three models all share a large part of the program, hence it was found convenient to build the entire system as a single program set ('EMFLOW') and allow choice of the required options by reading a controlling data file. The computational steps followed under the influence of different control options is shown in Fig. 4.2. The algorithmic path taken within the program can be seen to be arranged in a cascade or 'waterfall' fashion allowing the user to exit computations at any of the levels. A flow chart of the steps followed by the program when Routing I (the Spectral model) is chosen (by setting NFIELD, NFLOW, NTURB, NSCHEM = 1 and LREYN, NCHECK = 0) is shown in Fig. 4.3. A similar flow chart for the steps entailed in choosing Routing II (the  $\kappa - \epsilon$  model) is shown in Fig. 4.4. It may be noted here that the XI model has only one partial differential equation to solve while the  $\kappa - \epsilon$  model has two. The extra step in the XI model of computing  $\xi$  needs only the solution of an algebraic equation and that too only in an integral sense for the entire control volume.

A complete source code listing of all these routines is also provided for reference in Appendix F.

Provision is made for the program to auto-check whether the flow will indeed be turbulent or laminar by evaluating a Performance Index of Stirring (PI). This is believed to be a far more reliable estimate of the degree of turbulence than afforded by a simple estimation of overall Reynolds Number. The basis for the calculation of PI is described in Section 4.7.

#### **4.7 Some Useful Integral System Parameters**

The objective of this section is to extend concepts of turbulence theory to compute some overall system parameters which may be beneficial in attempting to answer practical questions such as;

- (1) Is the flow going to be laminar or turbulent?  
What is the degree of turbulence?
- (2) How long is it likely to take for an added contaminant to become homogeneously distributed in the melt?

##### **4.7.1 A Performance Index for Stirring**

To sustain turbulence in a flow system, it is necessary to have a steady flow of energy through the turbulent energy cascade. In an overall sense, (integrating over the contents of the whole vessel) the total rate of generation of turbulence should equal the total rate of dissipation of turbulence.

$$G = \Phi \quad (4.48)$$

It is, however, not necessary for the entire stirring power supplied to the system by the external agency (Electromagnetic force field, Gas injection or other as the case may be) to appear as turbulence<sup>(54,67-70)</sup>. Corrsin<sup>(67,68)</sup> based on Laufer's<sup>(54)</sup> measurements and later Brodkey<sup>(69,70,89)</sup> have postulated the existence of a factor of efficiency ( $\eta$ ) governing the generation of turbulence. Thus it may be said that:

$$\eta P = G = \Phi \quad (4.49)$$

The local generation term  $g_{ij}$  can be expressed as :

$$g_{ij}(r,x) = 2\mu_t \left[ \left( \frac{\partial U_x}{\partial x} \right)^2 + \left( \frac{\partial U_r}{\partial r} \right)^2 + \left( \frac{U_r}{r} \right)^2 + \frac{1}{2} \left( \frac{\partial U_r}{\partial x} + \frac{\partial U_x}{\partial r} \right)^2 \right] \quad (4.50)$$

evaluated at location  $(r,x)$ .

Integrating this over the whole volume gives:

$$G = \iiint_{Vol} g_{ij}(r,x) \, dVol \quad (4.51)$$

Taking a characteristic value for  $\mu_t$  as a constant over the whole flow field, the expression for the total generation, G may be written as :

$$G = 2\mu_t \iiint_{Vol} \left[ \left( \frac{\partial U_x}{\partial x} \right)^2 + \left( \frac{\partial U_r}{\partial r} \right)^2 + \left( \frac{U_r}{r} \right)^2 + \frac{1}{2} \left( \frac{\partial U_r}{\partial x} + \frac{\partial U_x}{\partial r} \right)^2 \right] dVol$$

$$P.I. = \frac{M_t}{\mu} = \eta P / \left\{ 2\mu \iiint_{Vol} \left[ \left( \frac{\partial U_x}{\partial x} \right)^2 + \left( \frac{\partial U_r}{\partial r} \right)^2 + \left( \frac{U_r}{r} \right)^2 + \frac{1}{2} \left( \frac{\partial U_r}{\partial x} + \frac{\partial U_x}{\partial r} \right)^2 \right] dVol \right\} \quad (4.52)$$

#### 4.7.2 Mixing Time Analysis

Following the analysis adopted by Corrsin<sup>(67)</sup> it may be shown that the time constant of mixing can be given by:

$$\tau = \frac{l^2}{12 D_c} \approx \frac{\lambda^2}{6\nu} \quad (4.53)$$

Following Taylor, the dissipation rate  $\phi$  may be expressed by

$$\phi \approx \frac{10}{3} \nu \frac{\bar{q}^2}{\lambda^2} = \eta \frac{P}{M} \quad (4.54)$$

using values averaged over the whole melt of mass M Kg.

In Isotropic turbulence at large Reynolds Numbers, Von Karman and Howarth deduced an approximate relation between dissipative scale  $\lambda$  and integral scale L

$$\frac{\lambda}{L} \sim \frac{1}{R_\lambda} \quad (4.55)$$

Using the constancy factor proposed by Corrsin we may rewrite the above relation as



$$\frac{\lambda}{L} = \frac{A}{R_\lambda}$$

81

(4.56)

$$R_\lambda = \frac{u' \lambda}{\nu} = \frac{q/\sqrt{3} \lambda}{\nu}$$

(4.57)

$$q \approx \frac{\sqrt{3} A L \nu}{\lambda^2}$$

(4.58)

substituting this in eqn.(4.53) gives:

$$\tau \approx \left\{ \frac{A^2}{21.6\eta} \frac{ML^2}{P} \right\}^{1/3}$$

(4.59)

where  $A \approx 20$  (approx.) as measured from behind grid turbulence measurements ( ). Expanding eqn.(4.59) gives

$$\tau \approx \left( \frac{A^2}{21.6\eta} \right)^{1/3} \left( L^2 \frac{M}{P} \right)^{0.333}$$

(4.60)

Defining 'mixing time' ( $t_m$ ) as the time required for concentration fluctuations to be within 5% variation from the mean, gives

$$t_m \approx 3 \tau$$

So

$$t_m \approx 3 \left( \frac{A^2}{21.6\eta} \right)^{1/3} \left( L^2 \frac{M}{P} \right)^{0.333}$$

(4.62)

If the same equation were written using power in Watts/Ton of melt ( $P_t$ ), the expression for  $t_m$  would read:

$$t_m \approx \frac{80}{\eta^{1/3}} \left( L^2 / P_t \right)^{0.333}$$

(4.63)

Sano and Mori<sup>(58)</sup> have found that measurements of  $t_m$  in various configurations and sizes of gas injected systems can be well correlated by an expression of the form:

$$t_m = 100 \left\{ \left( \frac{D^2}{H} \right)^2 / P_t \right\}^{0.337} \quad (4.64)$$

Fig. 5.33 taken from Sano and Mori<sup>(58)</sup> shows this correlation, together with the experimental results of various researchers working with gas injected systems. 'A' represents the correlation stated above. Lines B1 and B2 represent the experimental results of Haida et.al.<sup>(57)</sup> for a water model without and with slag respectively. Lines C1 and C2 represent the experimental results of Hsiao et.al.<sup>(56)</sup> for 60 ton and 6 ton gas injected ladles while line D represents the results of Lehrer<sup>(55)</sup> for a water model of a gas sparged system.

( Note here that the analysis of Sano and Mori is not being used but rather the fact that measurements quoted therein can be correlated by the expression above. The original analysis of Sano and Mori considered the time of mixing as determined by the circulation time for the mean flow in the vessel. However, as Nakanishi et. al.<sup>(59)</sup> have pointed out, mixing time in metallurgical melts appears to be dominated by diffusion (and eddy diffusion) rather than by bulk circulation time in the melt. This appears to be reasonable considering the high

Schmidt Numbers encountered in metallurgical flow systems. It should also be pointed out at this stage that the turbulence analysis followed in this work is quite general in nature and is not restricted in any way to gas injected systems. The value of turbulence efficiency would, of course, be expected to be different in different classes of systems.)

In keeping with the rather gross approximations made in this analysis, it may be written that:

$$L \approx D \approx H \approx L_v$$

in which case eqn.(4.63) becomes

$$t_m \approx \frac{80}{\eta^{1/3}} \left\{ L_v^2 / P_t \right\}^{0.333} \quad (4.66)$$

and eqn.(4.64) becomes:

$$t_m \approx 100 \left\{ L_v^2 / P_t \right\}^{0.337} \quad (4.67)$$

Equating the two gives:

$$\eta = (0.80)^3 = 0.51 \quad (4.68)$$

Somewhat surprisingly, this value of  $\eta$  for turbulence generation in gas injected ladles is found to match exactly with the value of 0.50 found by Laufer for pipe flow

turbulence generation and suggested by Corrsin as a first guess value for  $\eta$

This value of efficiency has been calculated for the case of gas injected ladle metallurgical systems. It is expected that the efficiency of turbulence generation would be different for different classes of systems. However in the absence of data of this sort for different systems of metallurgical interest, it is recommended that  $\eta = 0.50$  be used as a first guess until a better estimate is forthcoming.

Nakanishi et. al.<sup>(59)</sup> have noted in their work that the data they reviewed (see Fig.5.32) does not appear to demonstrate any dependance on the linear size of the particular equipment that was studied. Indeed at first glance this would appear to be true from Fig.5.32. However a closer look at the derived dependance of mixing time on linear size (Eq.4.66) indicated a relationship of the form

$$t_m \propto L_v^{(2/3)} \quad (4.69)$$

Now, in the data surveyed by Nakanishi et. al. the largest vessel is a 200 ton unit while the smallest is a 65 kg water model. This implies that the ratio:

$$(\text{max. vol. of vessel})/(\text{min. vol. of vessel})$$

is in the region of 340. Very roughly, this would indicate a ratio of characteristic linear dimensions to be around  $(340)^{(1/3)} = 7$

Now in eq. (4.69),  $L_v$ , the characteristic vessel dimension appears to the power of 0.66. This would mean that introducing into Fig. 5.32 a linear scale dependency as indicated by eqs. 4.66 and 4.67 would imply a change in Y-axis location of the data points by a factor of  $(7)^{(2/3)} = 3.6$  or, on the Log scale used in Fig. 5.32, a maximum difference of 0.5 units. This change is clearly of the same magnitude as the half decade variability of the data points themselves. It may therefore be inferred that an expression for mixing time such as the one given by eq. 4.66 does not necessarily contradict the findings of Nakanishi et. al. Indeed, the additional parameters of efficiency and linear vessel size may prove to be one of the reasons contributing to the spread in data points.

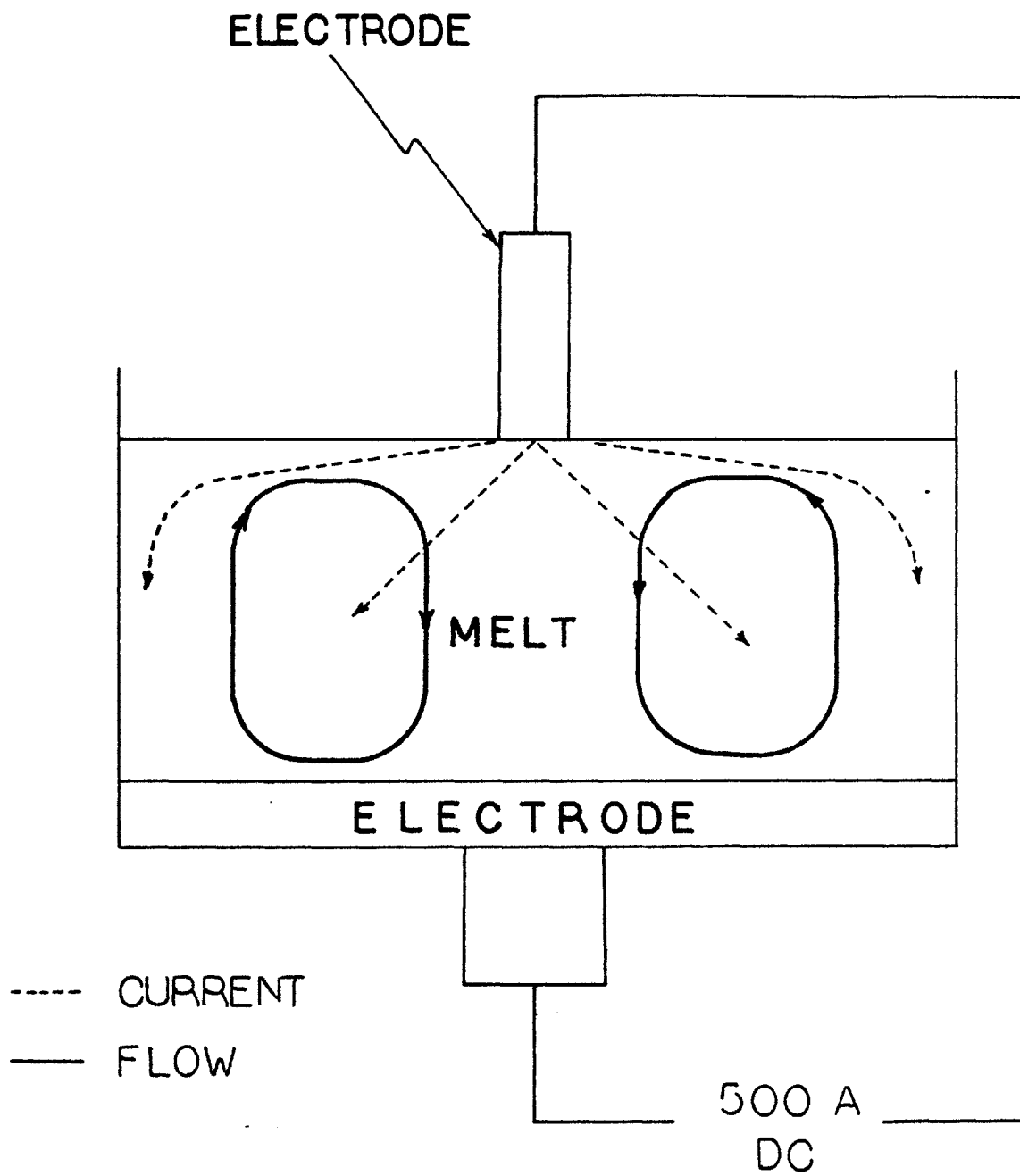


Fig. 4.1 Schematic of Flow System to be Modelled

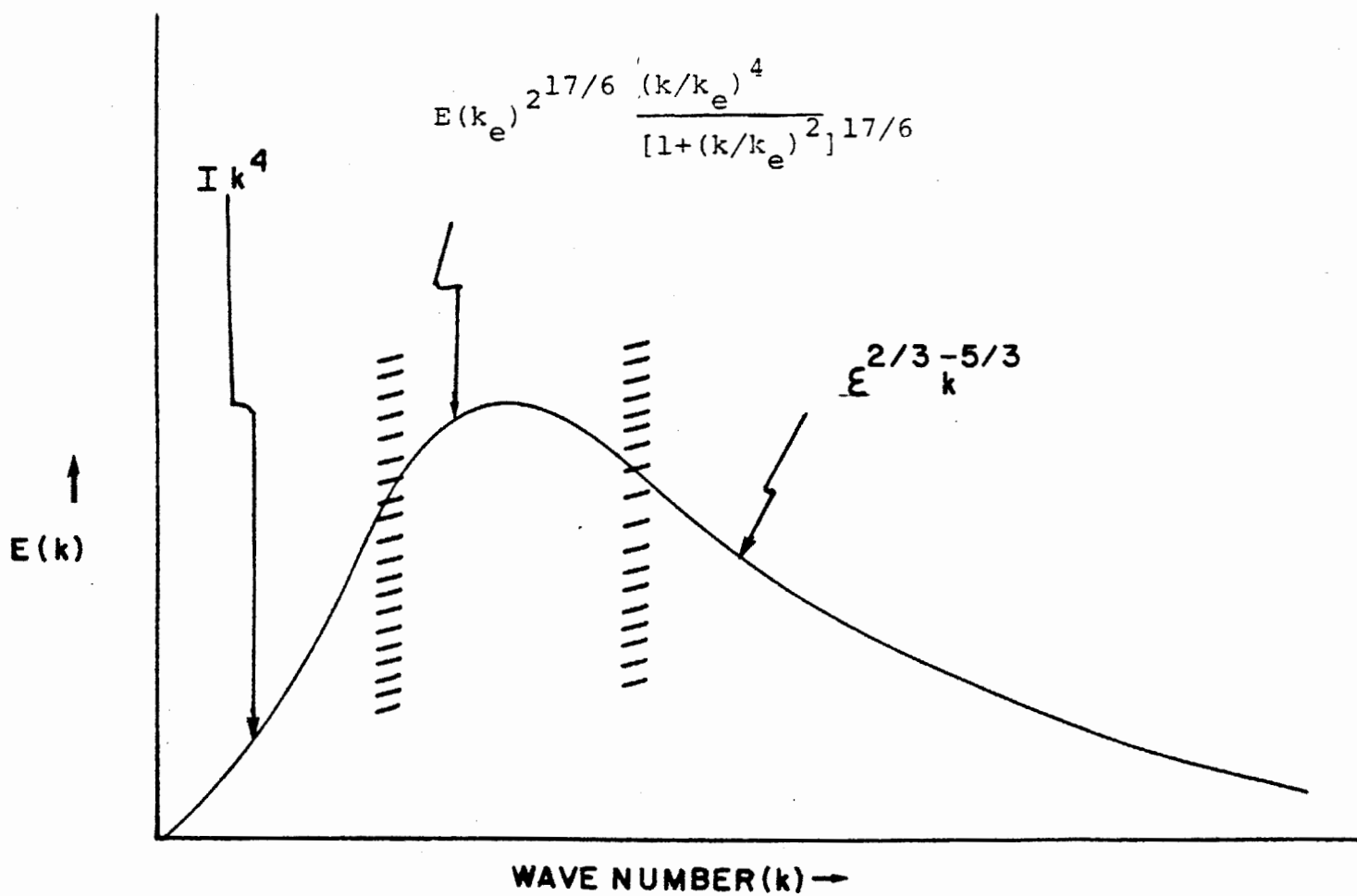
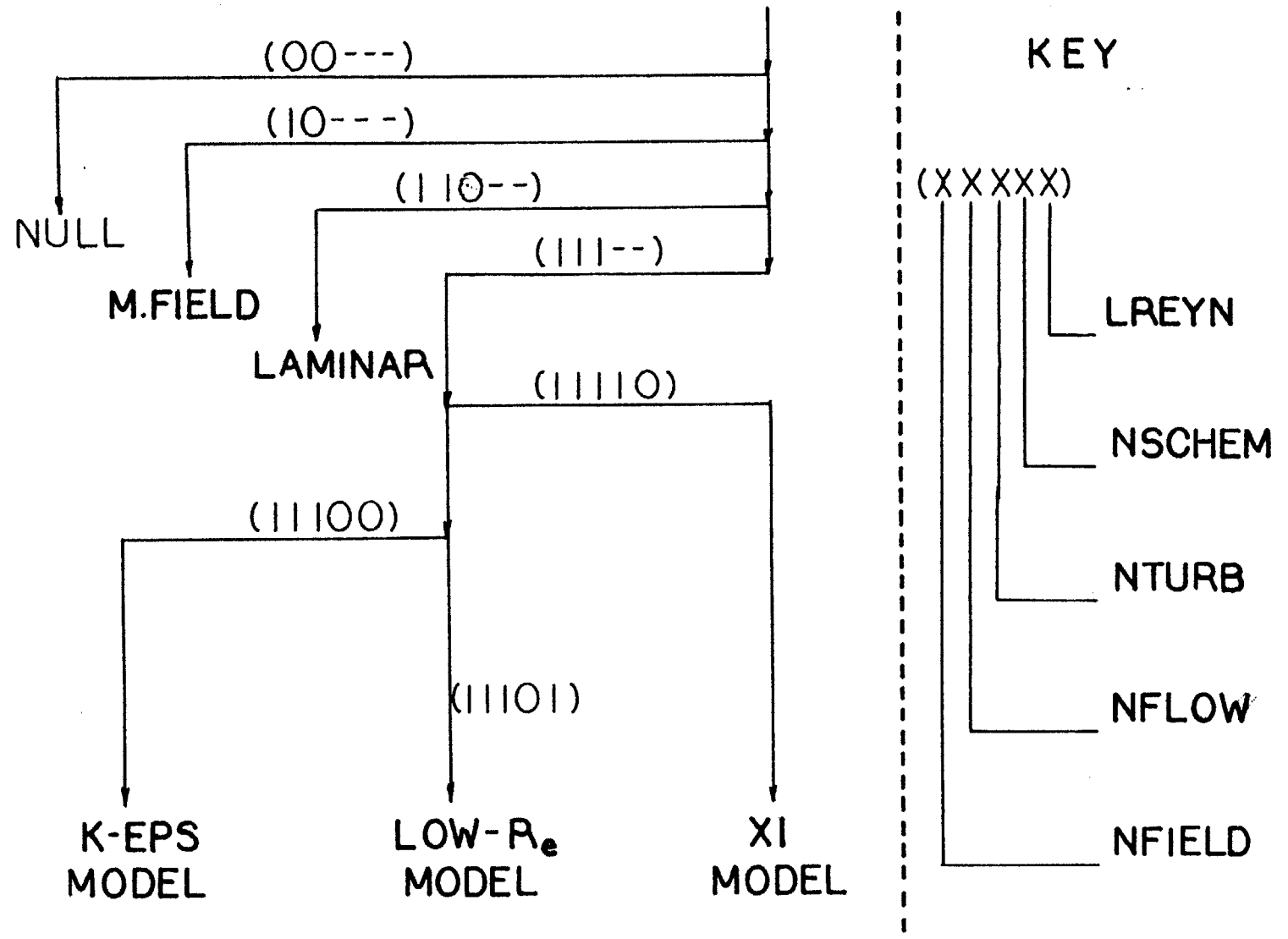


Fig. 4.2 Schematic of 3-D Turbulence Energy Spectrum

Fig. 4.3 Option Control in Computer Program





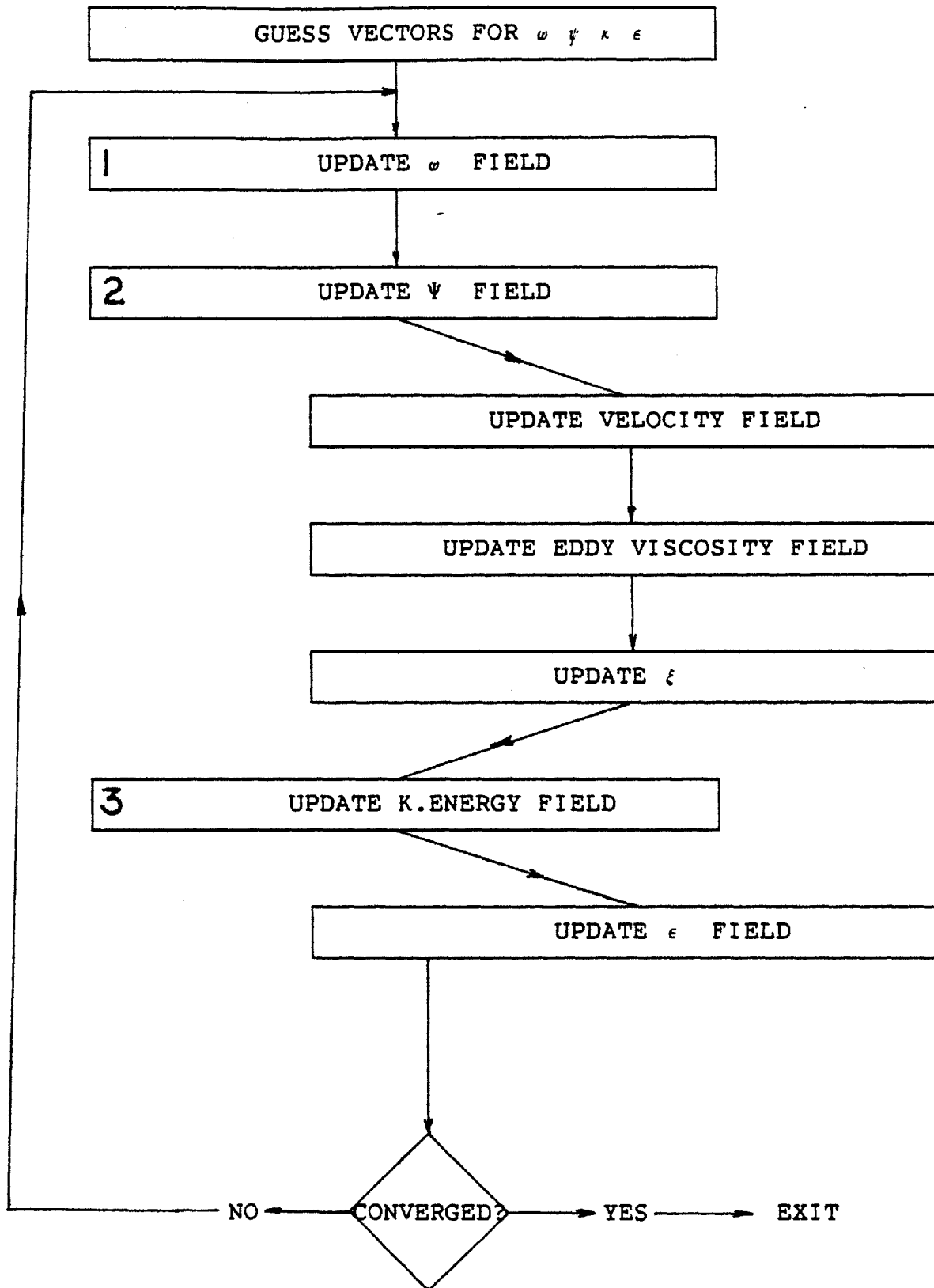


Fig. 4.4 Flow Chart of Calculations using XI Model (1 eqn. model)

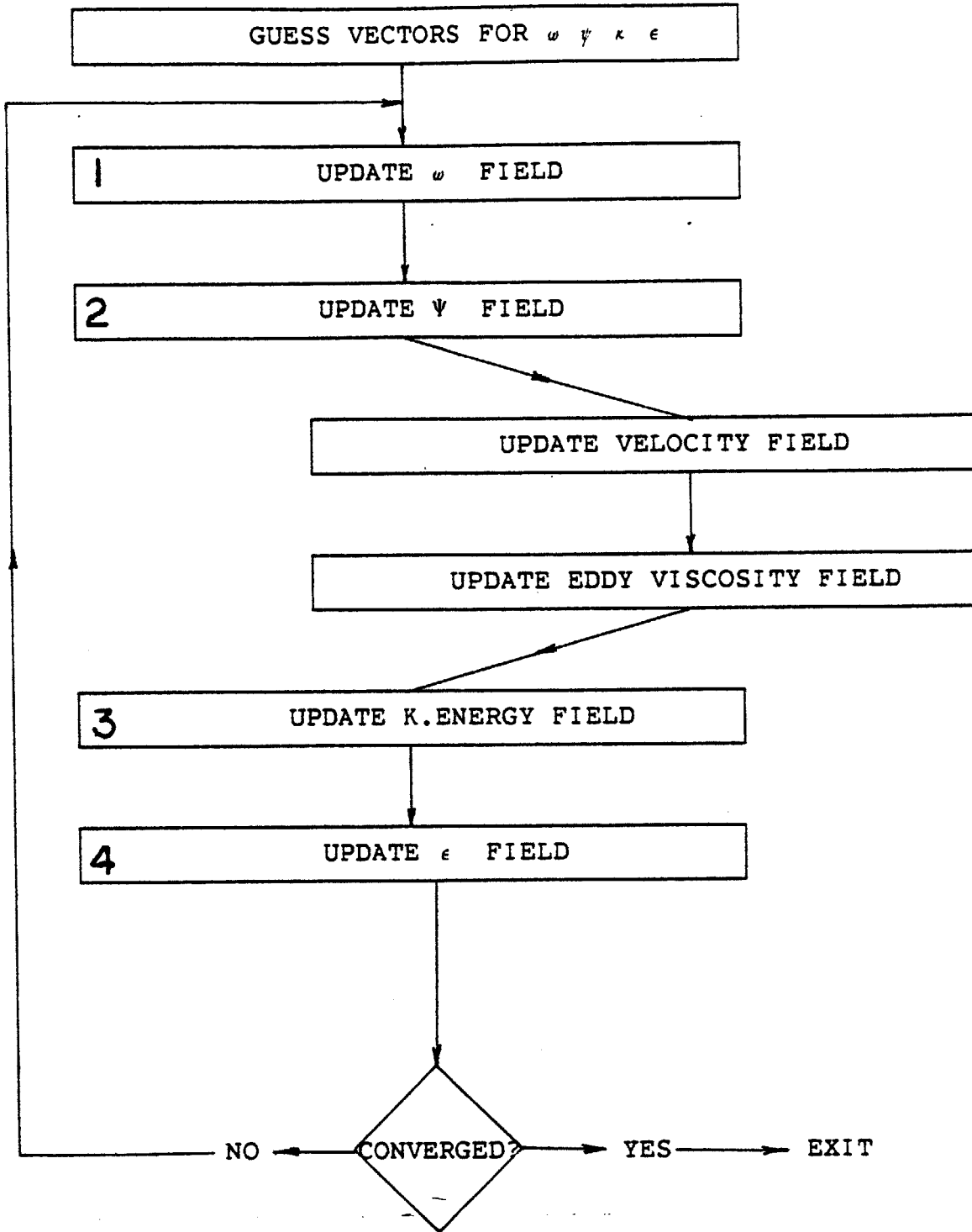


Fig. 4.5 Flow Chart of Calculations using  $\kappa$ - $\epsilon$  Model (2-eqn. model)

## 5 RESULTS AND DISCUSSION

In this chapter, results of experimental measurements and numerical computation will be compared and discussed. In addition the behaviour of the calculated values for certain practically relevant parameters like Performance Index (P.I. for short) and Mixing Time ( $t_m$ ) and finally the results of analyses pertaining to the use of thermal anemometry in liquid metal and also in the presence of magnetic fields will be presented. Throughout this chapter, Models I, II and III refer to the  $\kappa - \epsilon$  model, the Low Reynolds Number model and the XI model respectively.

### 5.1 Main Flow System - Measurements and Computations

Fig. 5.1 shows computed current density ( $J$ ) vector plot. Two facts stand out here: (1) Most of the current passes through in a region relatively close to the smaller electrode. This means that insulation of the wall of the vessel should not play too important a role as the leakage wall current may, quite reasonably, be expected to be small. (2) The current divergence is the strongest in the immediate neighborhood of the small electrode. Hence this region is crucial in driving the overall flow.

Fig. 5.2 shows a computed velocity vector plot. The formation of one circulation loop is clearly seen. The arrows denote the velocity vector at the center of the arrow. The broken arrows, however, indicate only the local flow direction, the magnitude being greater than the maximum for

the plot (3.0 cm/sec). Fig. 5.3 through 5.9 show the distribution of the local velocity across the radius at various depths. The absolute magnitude velocity was chosen as a basis for presentation since the hot film probe in the experiment actually measured the absolute resultant velocity. The good agreement of the experimentally measured velocities with the computed values is evident. Also to be noticed is the fact that all three models ( I-  $\kappa - \epsilon$  , II- Low Reynolds Number version of  $\kappa - \epsilon$  , and III- XI model) predict exactly the same distribution except at the centerline. This extremely close correspondence between the calculations is partly because the flow is largely convection driven and partly because the turbulent viscosity predictions also match very closely (see Fig. 5.24). Fig. 5.8 clearly shows up the location of the eye of the vortex at about 5 cms radius. Notice also that the dip in the velocity near the vortical region is also reflected in the experimental measurements.

Fig. 5.35 shows a typical experimentally measured spectrum of turbulence with the various pertinent operating conditions and directly computed parameters noted on it.

Scalar valued overall Dissipation Energy is plotted along the radial direction at different depths in Fig. 5.10 through 5.16. Once again two facts are to be noticed here:

- (1) The XI model, even though it is a one equation model (see description in chapter 4) and therefore not containing

any of the constants involved in the  $\epsilon$  transport equation, comes remarkably close to the predictions using the  $\kappa - \epsilon$  model of turbulence. The Jones and Launder Low Reynolds Number model, however, also predicts approximately the same as the other two models being only marginally different from the  $\kappa - \epsilon$  model. (2) All three models seem to predict the experimentally observed values of dissipation energy reasonably well in general. The agreement gets progressively worse as one approaches the depth at which the vortical point is located and gets better the farther away one is from the vortex.

Figures 5.17 through 5.23 show the variation of local RMS turbulence velocity in the form of radial scans at various depths from the melt surface. Once again the new XI model predicts values of the RMS fluctuations very close to those predicted by the  $\kappa - \epsilon$  model. Comparison with experimentally measured values is however less encouraging. In general the numerical calculations overpredict the values of the fluctuating velocity. This may seem somewhat surprising, in view of the fact values of dissipation energy were really quite well represented by the computations. One possible reason for this apparently paradoxical behaviour may be related to the fact that the major contribution to the total dissipation energy comes from a range of frequencies towards the higher frequency end of the spectrum while the major

contribution to the value of  $\kappa$  comes from frequencies in the lower ranges, particularly in the vicinity of the spectral density peak. Now the spectrum analysis equipment used in this experimentation ( as in most cases ) uses an AC coupled mode with a lower frequency cut off at 0.1 Hz (which is used by the system to distinguish the fluctuating from the mean component). This low frequency truncation would result in lower values for experimentally measured  $\kappa$  but would not affect the value for  $\epsilon$ . Also, the higher wavenumber region of the spectrum has been shown to be closer to isotropy than the low wavenumber regions. Since the dissipation energy is a high wavenumber phenomenon it is less likely to be affected by the assumption of isotropy than the kinetic energy and RMS fluctuation, which have a major contribution from the low wavenumber range of the spectrum. It would, however, seem hard to attribute the discrepancy entirely to these effects and it is likely that there is a combination of factors at play here.

Fig. 5.24 is a representative plot of the variation of effective viscosity across the radius. As may be expected, the three models predict almost the same values of turbulence enhanced viscosity, though the XI model shows a peak at the centerline while the  $\kappa - \epsilon$  model shows a peak somewhat away from the centerline. The wall neighbor node has exactly the same value with all three models since it is controlled

largely by the 'Wall Function' calculation scheme which is common to all three models.

The concept of the efficiency of turbulence generation plays a central role in the formulation of the XI model. On the basis of the considerations discussed in Chapter 4, a value of  $\eta = 0.5$  was used in all the preceding calculations. Fig. 5.25 and Fig. 5.26 show the effect of varying the value of  $\eta$  on the  $\epsilon$  and RMS fluctuating velocity profiles across the radius of the melt. Decreasing the value of  $\eta$  lowers the calculated value of both the Kinetic and Dissipation energies all over the field, or in general, decreases the degree of turbulence in the melt.

Figures 5.27, 5.28 and 5.29 show contour plots for Kinetic Energy, Dissipation Energy and Effective Viscosity from calculations using the XI model. These plots present the same information as in the radial plots in a more compact fashion and are included here mostly to provide a 'bird's eye view' of the system.

## 5.2 Performance Index Computations

Fig. 5.30 shows sample voltage outputs from the anemometer when different currents are passed through the melt. The 100 Amp. trace could be said to be laminar or at least substantially so. The 250 Amp. trace shows up what is probably the onset of turbulence and is what may be

characterised as 'weakly turbulent'. The 500 Amp. setting appears definitely turbulent. The transition from laminar to turbulent flow occurs somewhere between 100 A and 250 A possibly close to the 250 A end. Even a qualitative understanding of the onset of turbulence can be important because, as will be discussed later in this chapter, the rate of turbulence dissipation can be an important parameter in determining mixing, homogenisation and even reaction rates in the melt. Use of vessel Reynolds Number to predict the onset of turbulence may be misleading. For example, Reynolds Numbers based on the depth of melt and centerline velocity for the 100, 250 and 500 A cases are:

$$R_{100} = 2090 \quad R_{250} = 5570 \quad R_{500} = 11146$$

It would be difficult, if not impossible, to judge the onset of turbulence without a priori knowledge of the dynamics of the specific flow configuration ( an example would be the transition point for pipe flow which is placed around  $2300 = Re_d$  .) This would be particularly difficult for the more complex case of a recirculating flow system.

The whole concept of Performance Index (P.I.) fits in here. The value of PI is an immediate intuitive estimate of the 'degree of turbulence' in the whole flow field and reflects the ratio of Turbulent viscosity to Molecular Viscosity in an integral sense. Hence a value of  $PI=0$  would indicate an ideally laminar flow situation. Intuitively, it



may be said that any flow with a value of PI greater than 10 can be considered turbulent. The values of PI for the three current settings are:

$$PI_{100} = 5.5 \quad PI_{250} = 12.2 \quad PI_{500} = 23$$

Looking at these figures one can place the transition to turbulence as somewhere between 100 A and 250 A, possibly closer to 250 A. This is similar to what may be deduced by examining the traces in Fig. 5.30.

Fig. 5.31 shows a plot of the variation of PI with current. Such a plot could prove useful in predicting the degree of turbulence in a melt as the operating parameter (current) is changed.

### 5.3 Mixing Time Computations

Fig. 5.32 is the plot due to Nakanishi and Szekely<sup>(59)</sup>, relating mixing time to specific power supplied to the melt (Total power/ mass of the melt) for various metallurgical systems and is excerpted from Reference (60). This represents the first attempt to relate metallurgical system mixer performance to turbulence parameters and is placed here in context of what is to follow. Fig. 5.32 is taken from Sano and Mori<sup>(58)</sup>. Using a value of turbulence efficiency ( $\eta$ ) as 0.5 and using eq.4.63 to calculate mixing time, values of  $t_m$  calculated for the experimental setup of this work (with currents = 100, 250, 500 and 2500 Amperes) have also been

indicated on the figure with stars. Line A represents the correlation stated in equation 4.64. Lines B1 and B2 represent the experimental results of Haida et. al.<sup>(57)</sup> for a water model without and with slag respectively. Lines C1 and C2 represent the experimental results of Hsiao et. al.<sup>(56)</sup> for 60 ton and 6 ton gas injected ladles while line D represents the results of Lehrer<sup>(55)</sup> for a water model of a gas sparged system.

Fig. 5.34 shows the variation of overall mixing time for the current stirred melt used in this work, computed using the XI model with  $\eta = 0.5$ . Thus the melt could be expected to be 95% mixed in 24 seconds with a 500 Ampere current while it would take 120 seconds with a 100 A current and 4.8 seconds with a 2500 A current. Such a plot could be used in industrial practice to determine how long a blow or current stirring should be applied in order to get good mixing in the melt.

#### 5.4 M.H.D. Flow around a cylinder

Figs. 5.36 through 5.41 relate to the behaviour of a heated cylinder subjected to a cross flow of metal and a magnetic field on the orthogonal axes. Fig. 5.36 shows the schematic of the flow situation. Figs. 5.37 and 5.38 are presented to serve as a check on the computational technique used in this work. Fig. 5.37 shows a comparison of calculated

value of Drag Coefficient at zero magnetic field with corresponding values found by other workers. Fig. 5.38 compares the calculated values of Nusselt Number with the measurements of Ishiguro et. al.<sup>(95)</sup> and the older computations of Grosh and Cess<sup>(96)</sup>. The good agreement in both figures is evident.

Fig. 5.39 shows the computed variation of overall Nusselt Number with Magnetic Interaction (N). It is clear that even substantial levels of magnetic interaction would affect the heat transfer rate only marginally.

Fig. 5.40 shows, in a similar fashion, the corresponding variation of overall Sherwood Number for mass transfer from the cylinder. This shows a considerably stronger dependence on the magnetic interaction, the strongest dependency being demonstrated by the drag coefficient on magnetic interaction (Fig. 5.41). This last result would indicate the need for caution when interpreting results obtained by using drag probes to measure flow fields in the presence of magnetic fields. Also shown in this last figure are the results of Kalis et. al.<sup>(87)</sup>. The discrepancy between the two has been shown to be probably due to the use of a rather coarse computational grid near the cylinder wall.

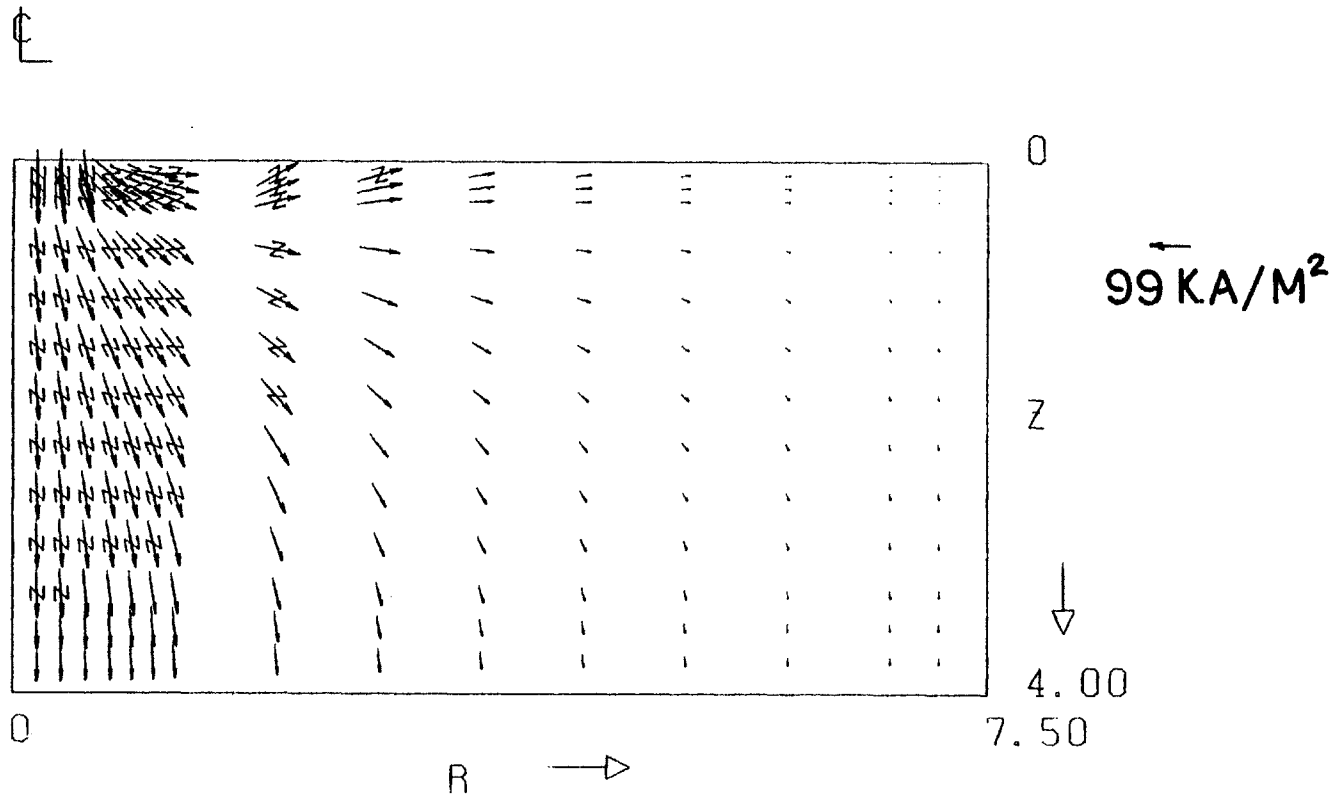
## 5.5 Calibration Drift Correction

Figs. 5.42 and 5.43 relate to typical results obtained from the calibration drift correction routine described in Chapter 4. Fig. 5.42 shows a typical calibration drift if plotted using as-measured values of voltage increment against flow velocity. Fig. 5.43 shows the same basic data but after the voltages have been adjusted to correct for drift. The improved reproducibility of the corrected calibration is evident.

## 5.6 Summary

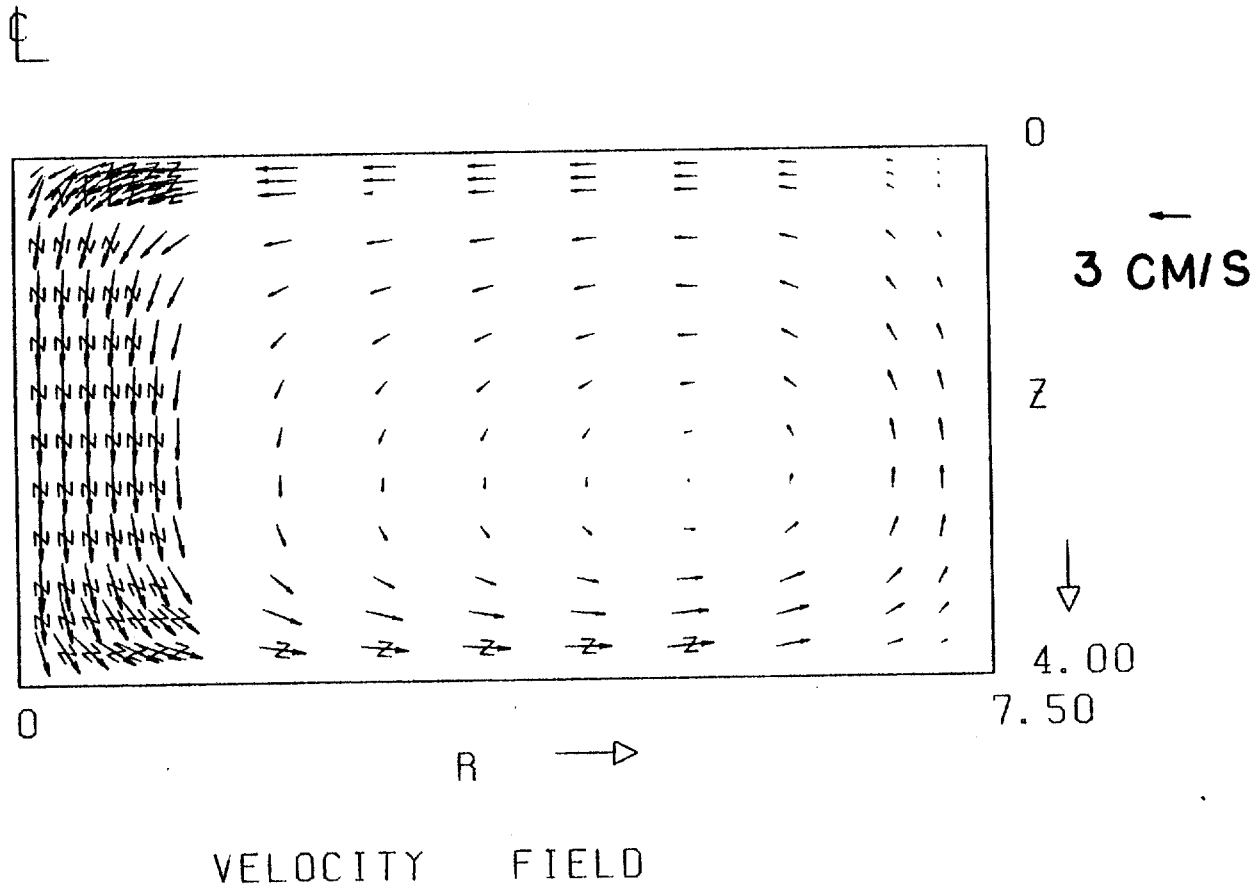
To summarise, it may be said that the  $\kappa - \epsilon$  model as well as the XI model predict mean flow quite well and turbulence dissipation rates only reasonably well but RMS fluctuating velocities are not as well predicted. The MHD flow analysis, though restricted to orthogonal fields, indicates negligible effects of magnetic fields on anemometer behaviour in fields of the range found in this work. The PI criterion has been shown to predict the onset of turbulence reasonably well, though somewhat constrained in its use by the need to specify a value for  $\eta$ . The calibration drift correction scheme has been shown to improve the reproducibility of calibration curves considerably. The theory proposed for turbulent mixing in metallurgical melts appears to correlate experimental and plant scale data reasonably well.

Fig. 5.1 Computed Current Density Vector Plot



CURRENT DENSITY FIELD

Fig. 5.2 Computed Velocity Vector Plot



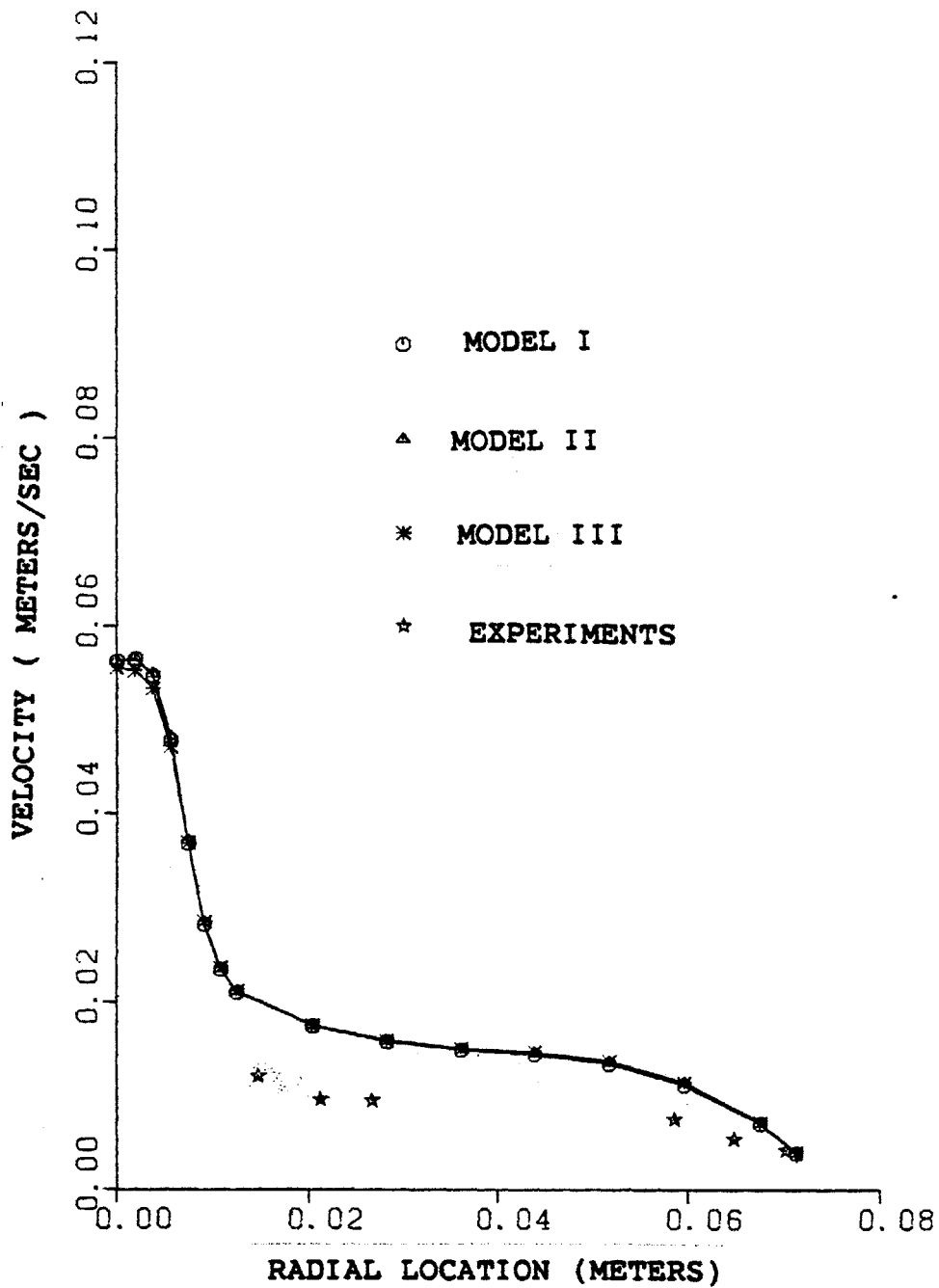


Fig. 5.3 Magnitude of Velocity, Radial Scan - Depth = 0.7  
cms

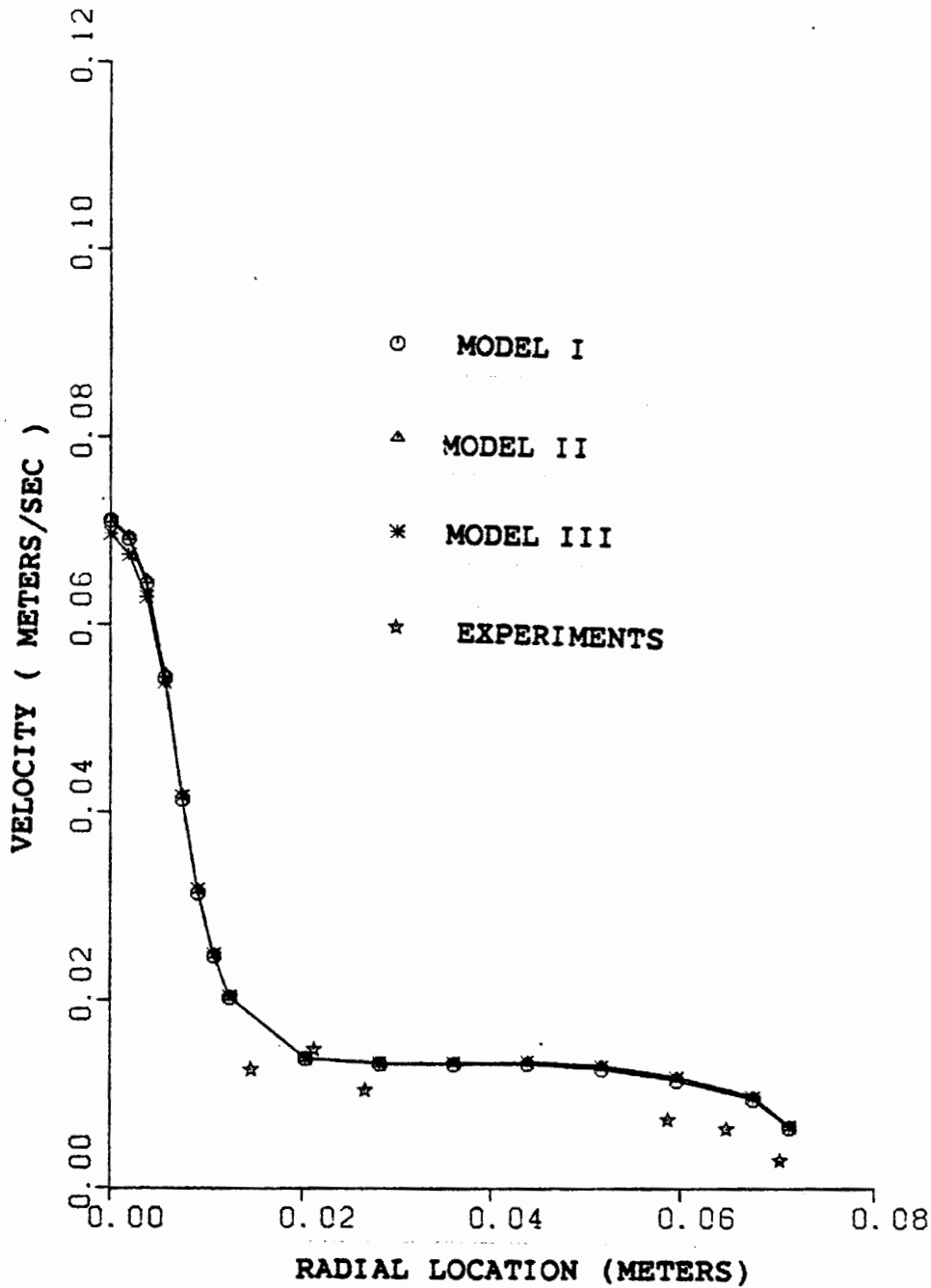


Fig. 5.4 Magnitude of Velocity, Radial Scan - Depth = 1.05  
cms



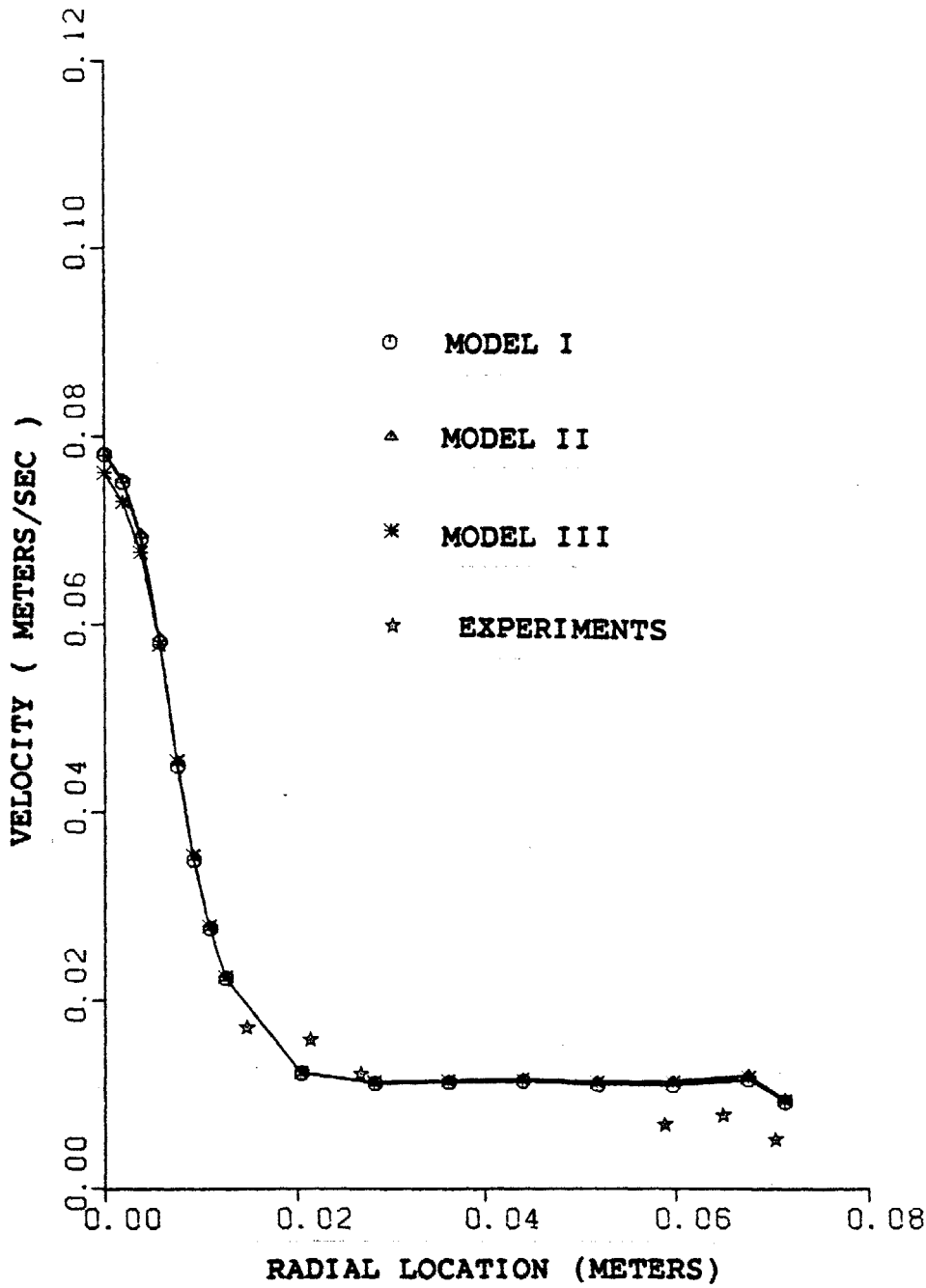


Fig. 5.5 Magnitude of Velocity, Radial Scan - Depth = 1.44 cms

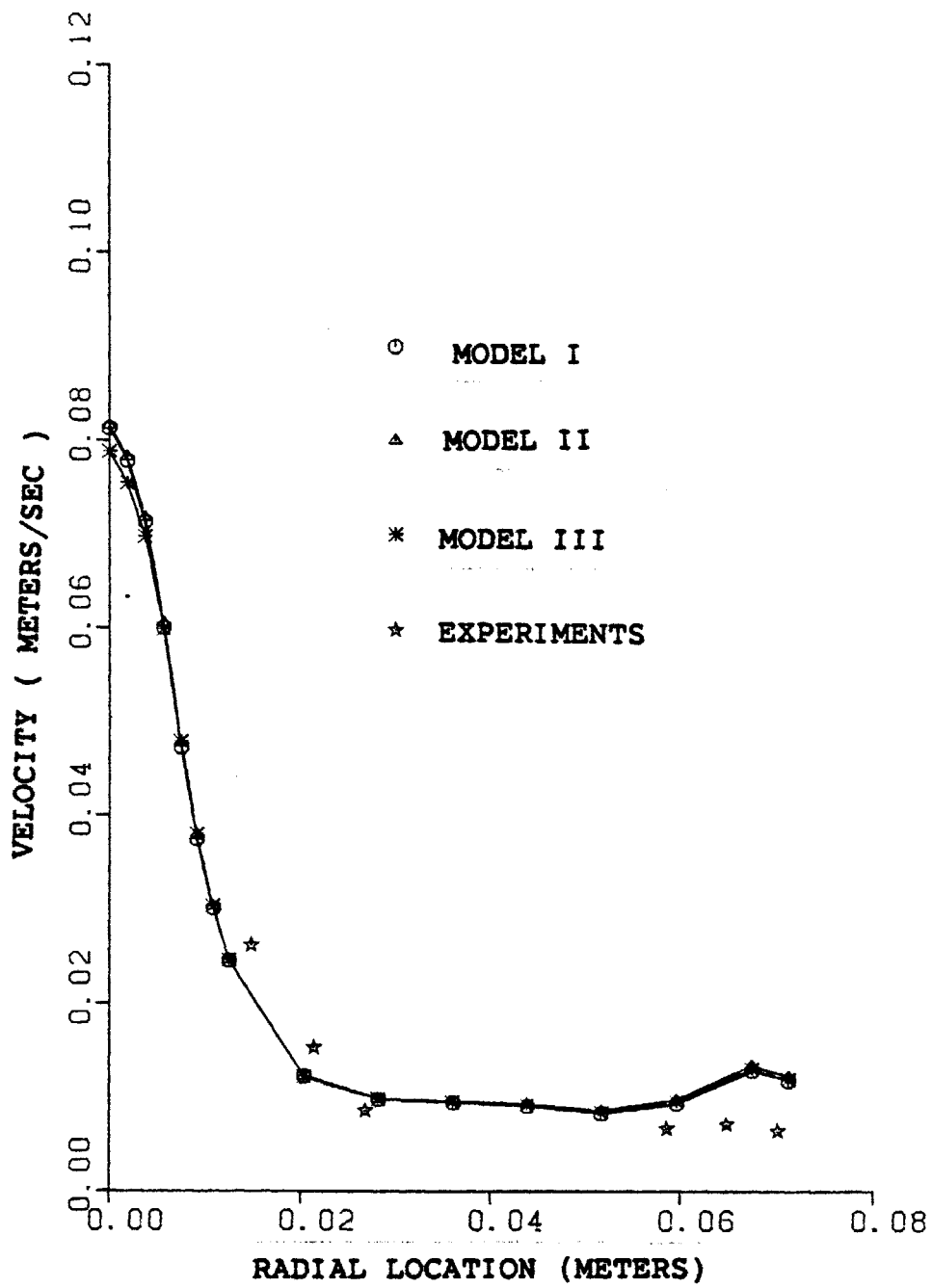


Fig. 5.6 Magnitude of Velocity, Radial Scan - Depth = 1.79 cms

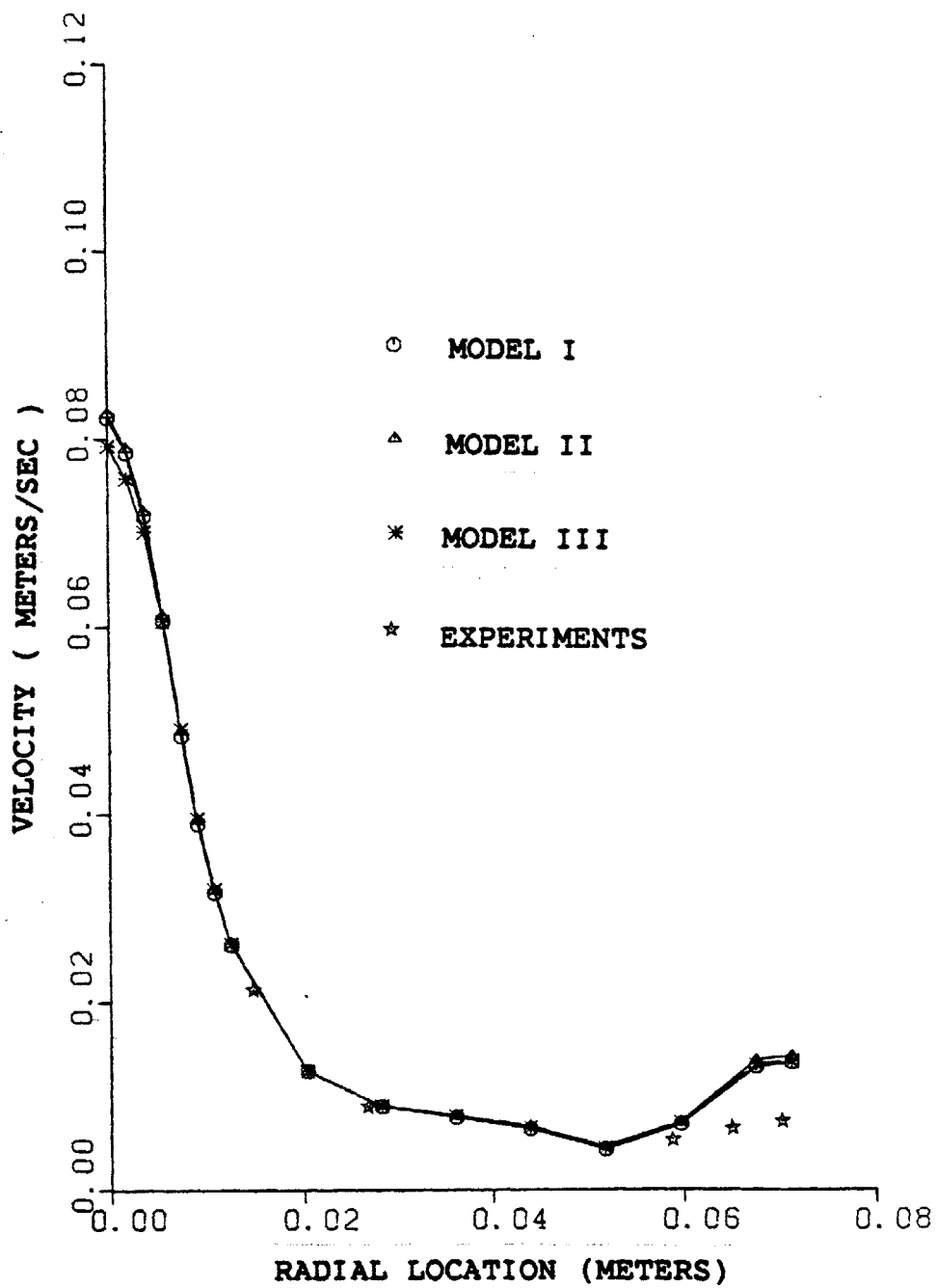


Fig. 5.7 Magnitude of Velocity, Radial Scan - Depth = 2.14 cms

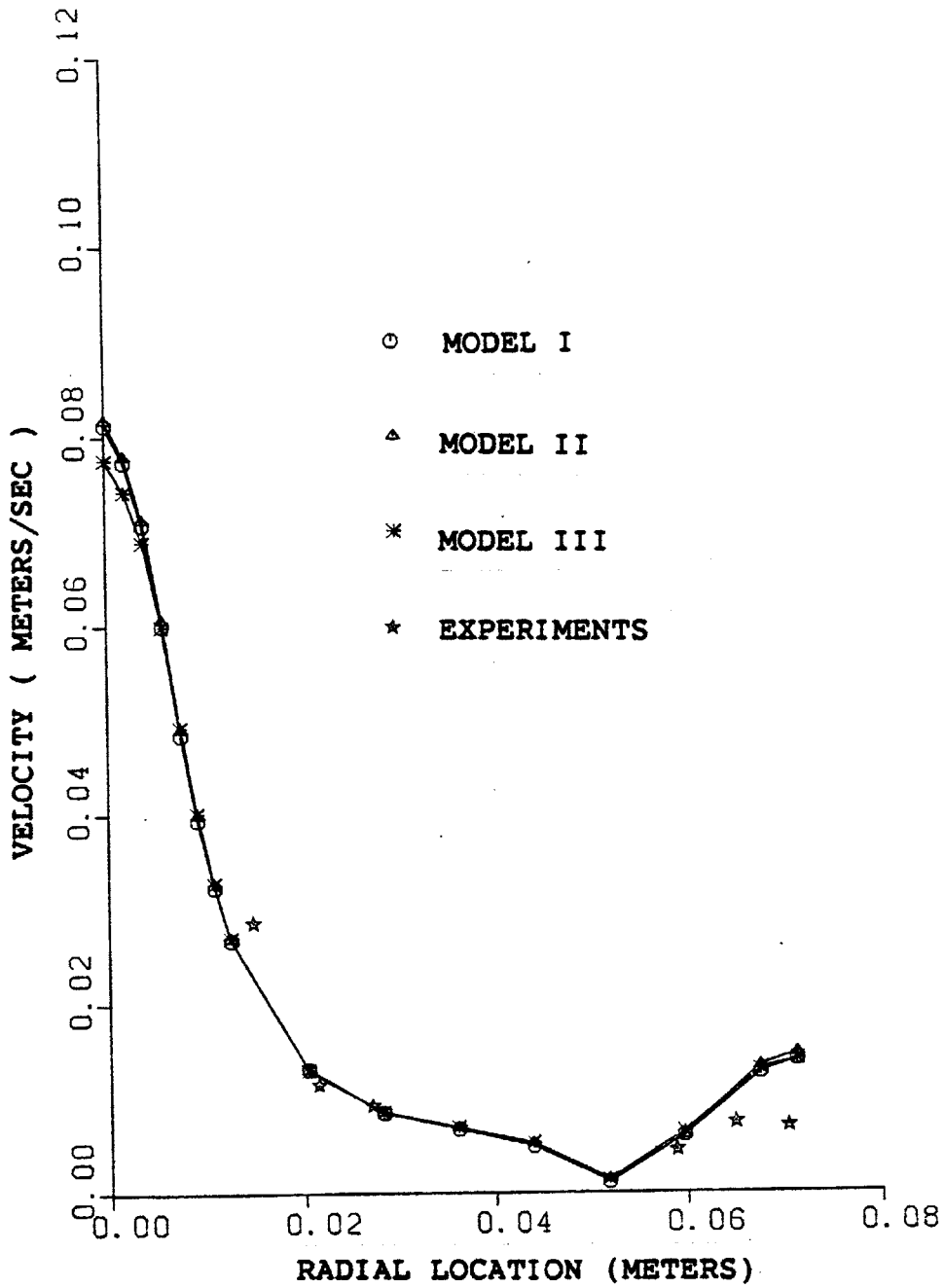


Fig. 5.8 Magnitude of velocity, Radial Scan - Depth = 2.5  
CMS

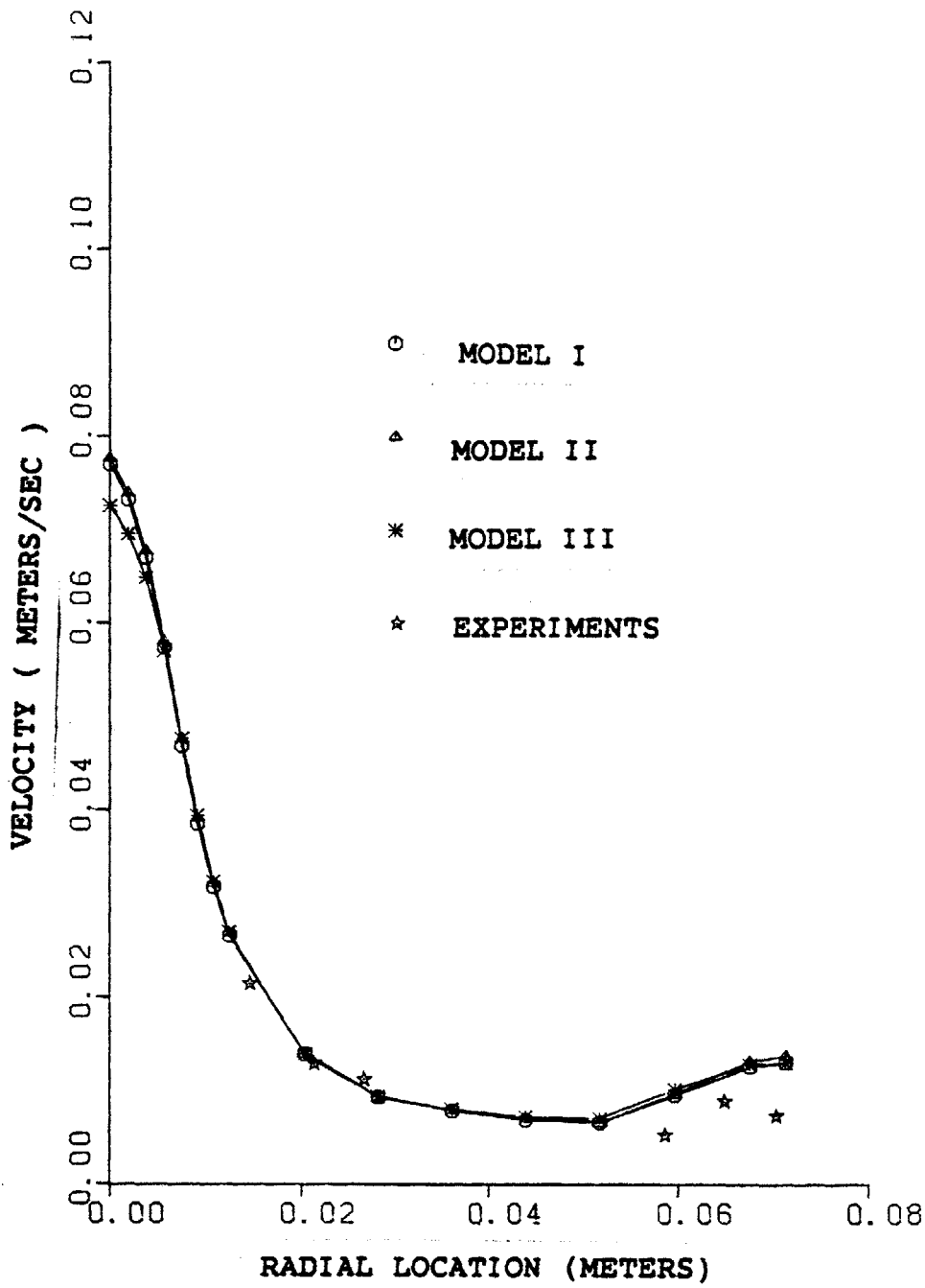


Fig. 5.9 Magnitude of velocity, Radial Scan - Depth = 2.85 cms

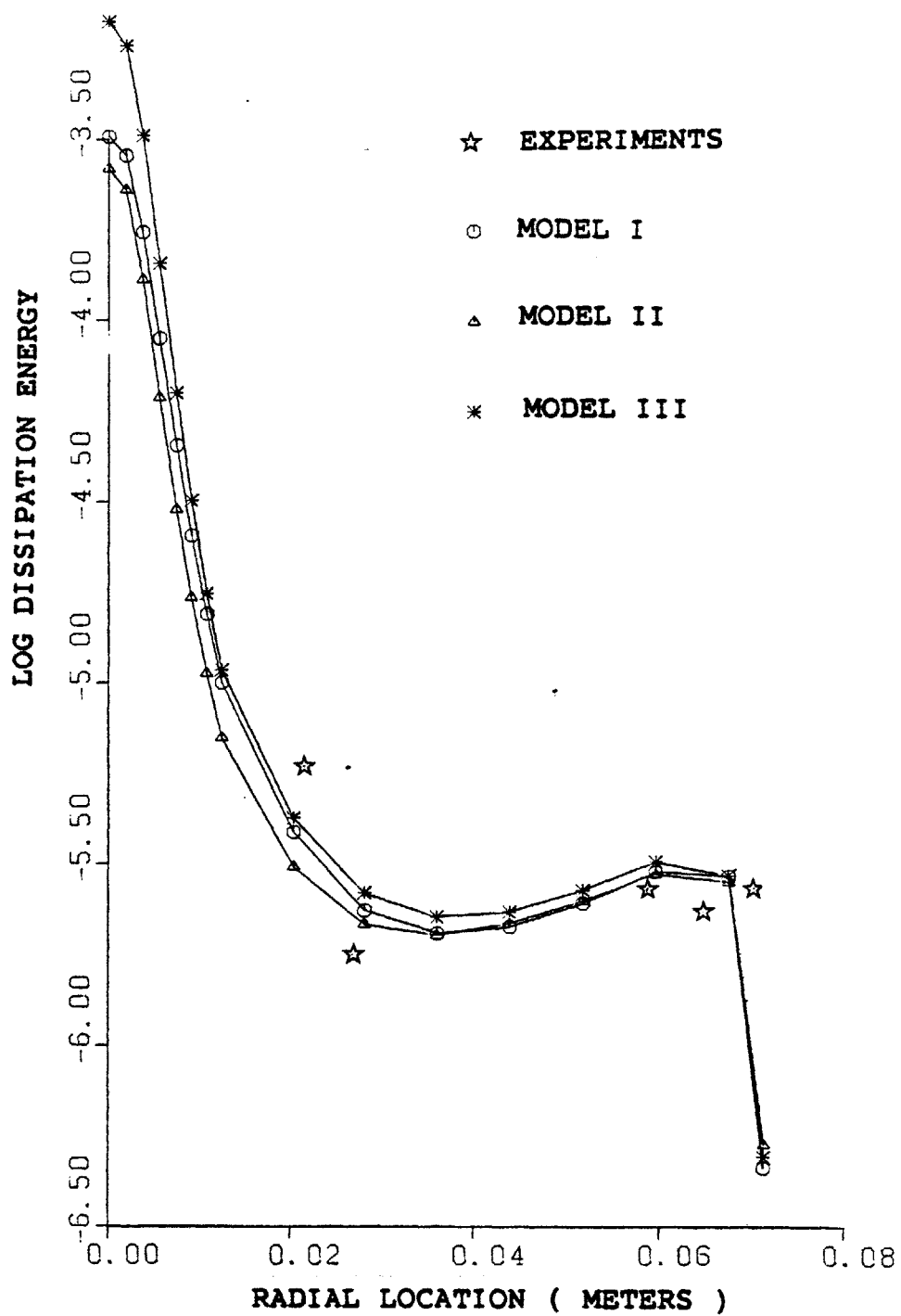


Fig. 5.10 Turbulence Dissipation Energy (W/Kg.s) Depth = 0.70 cms

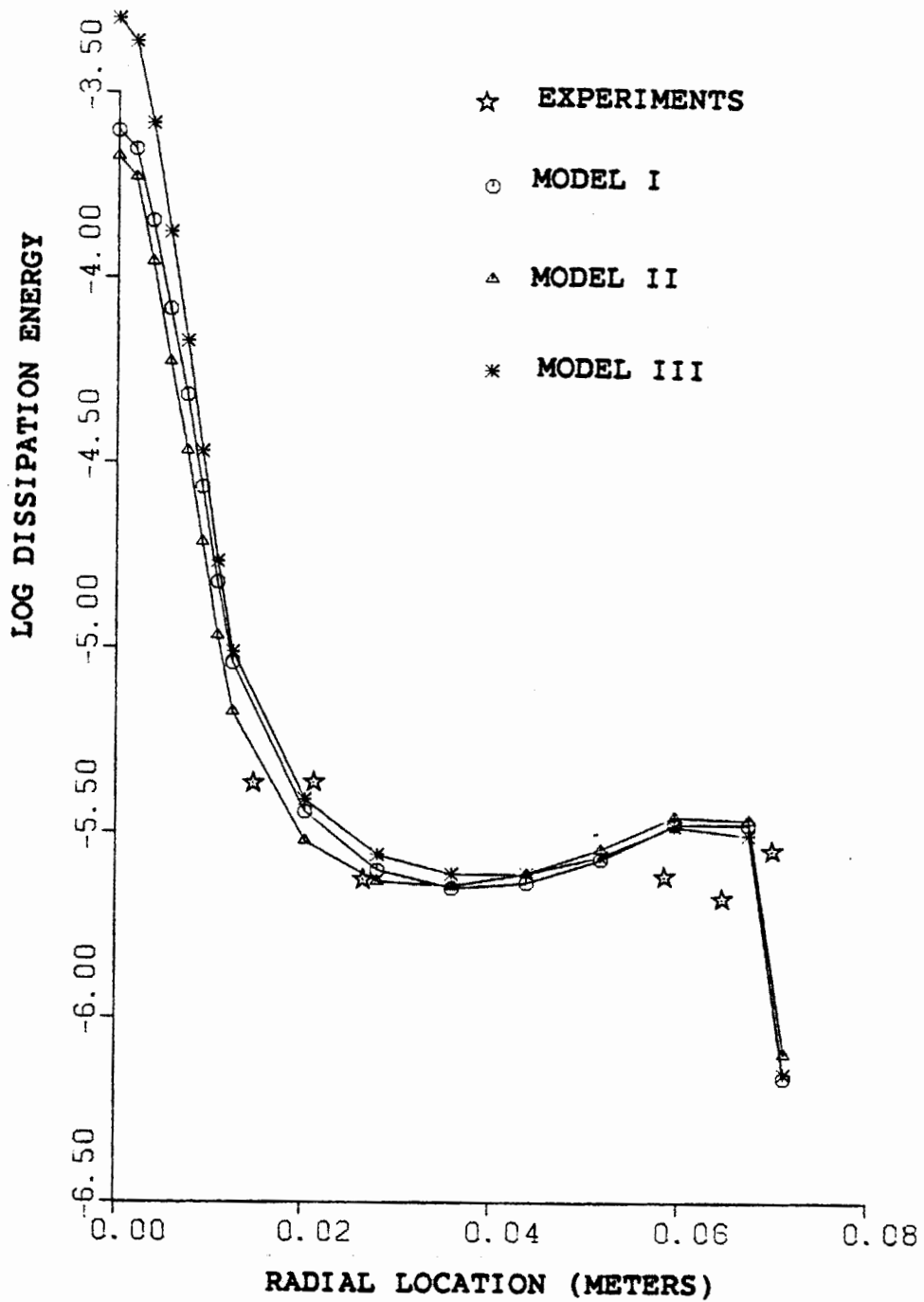


Fig. 5.11 Turbulence Dissipation Energy (W/Kg.s) Depth = 1.05 cms

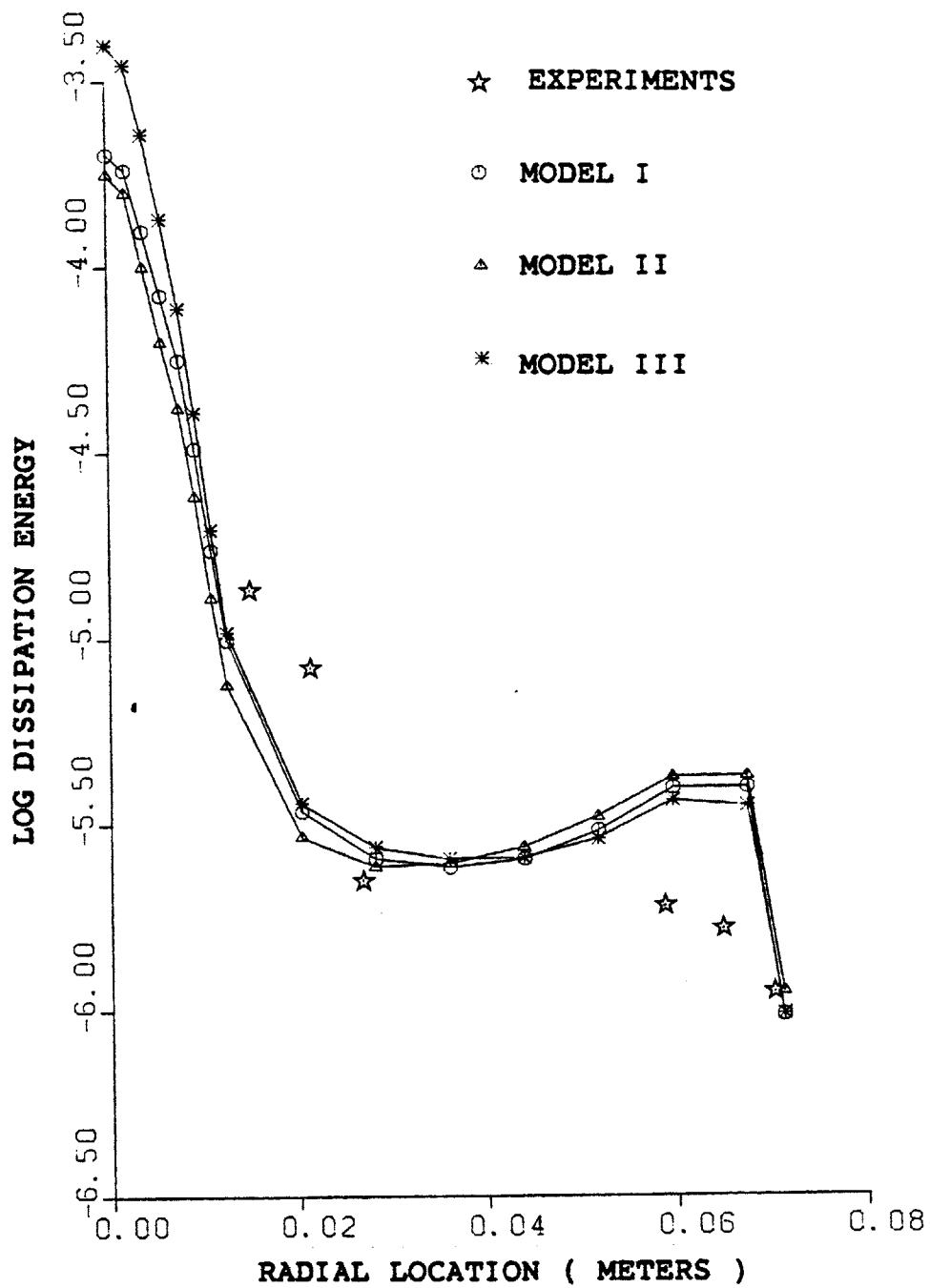


Fig. 5.12 Turbulence Dissipation Energy (W/Kg.s) Depth = 1.44 cms



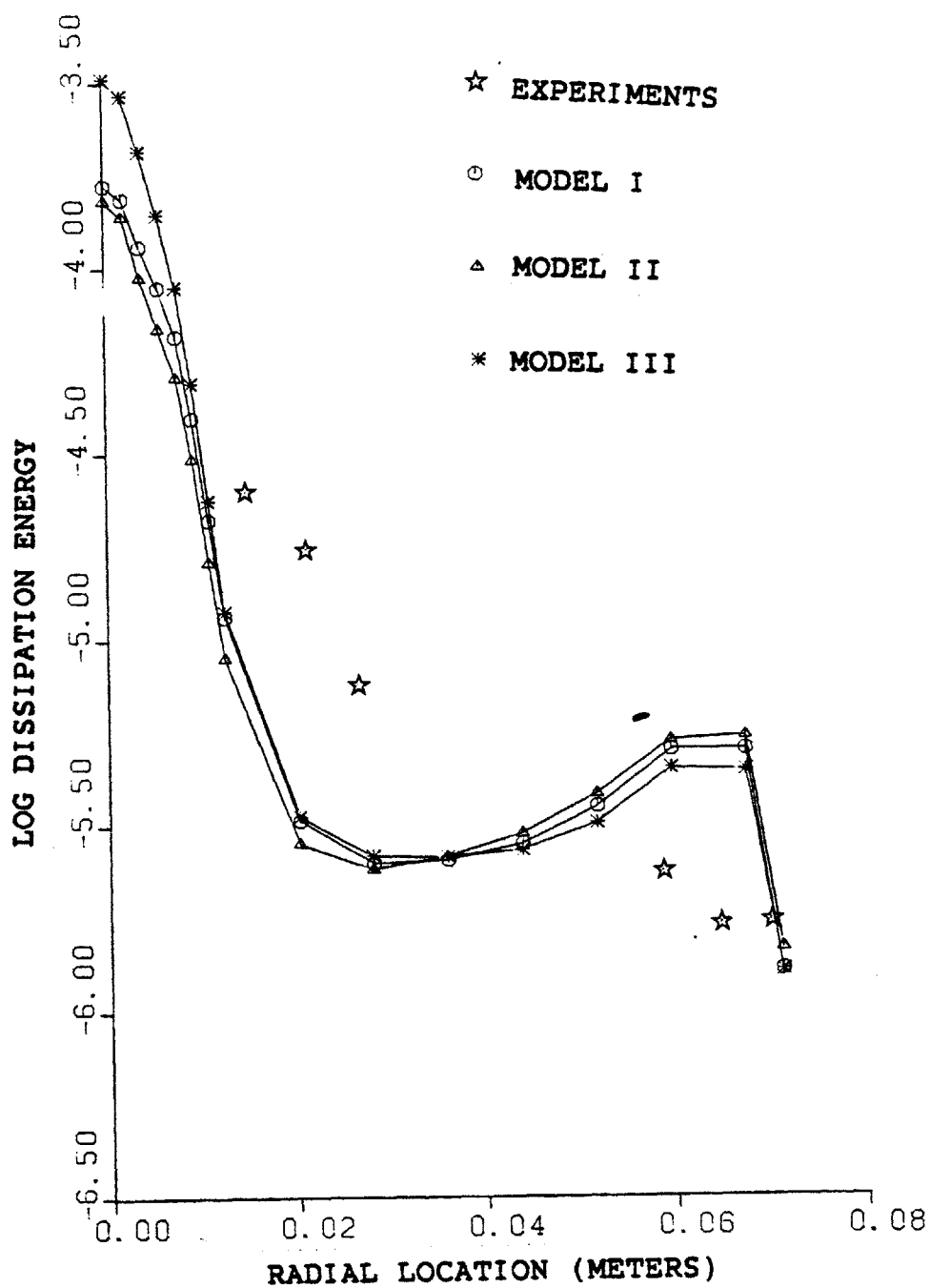


Fig. 5.13 Turbulence Dissipation Energy (W/Kg.s) Depth = 1.79 cms

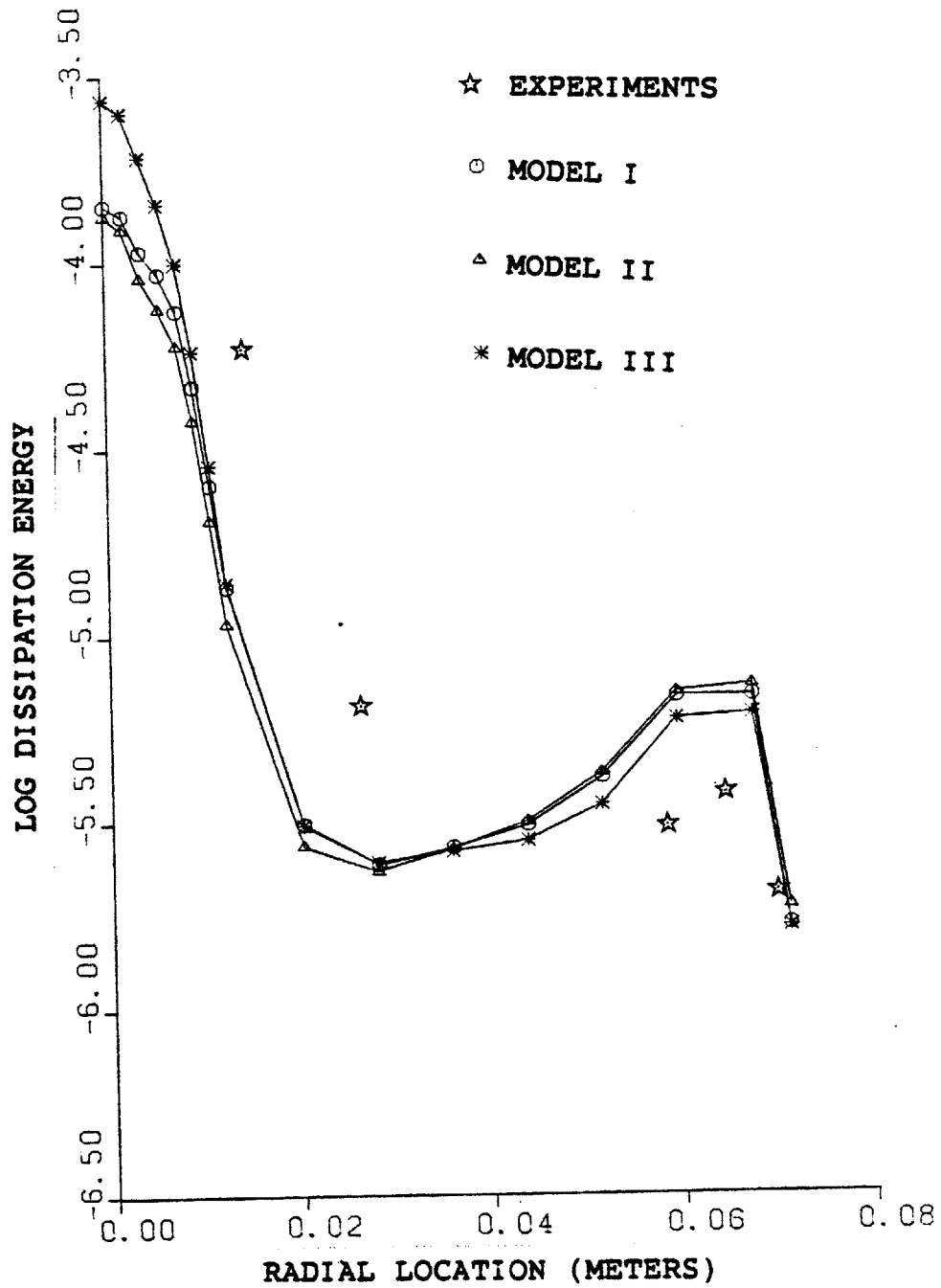


Fig. 5.14 Turbulence Dissipation Energy (W/Kg.s) Depth = 2.14 cms

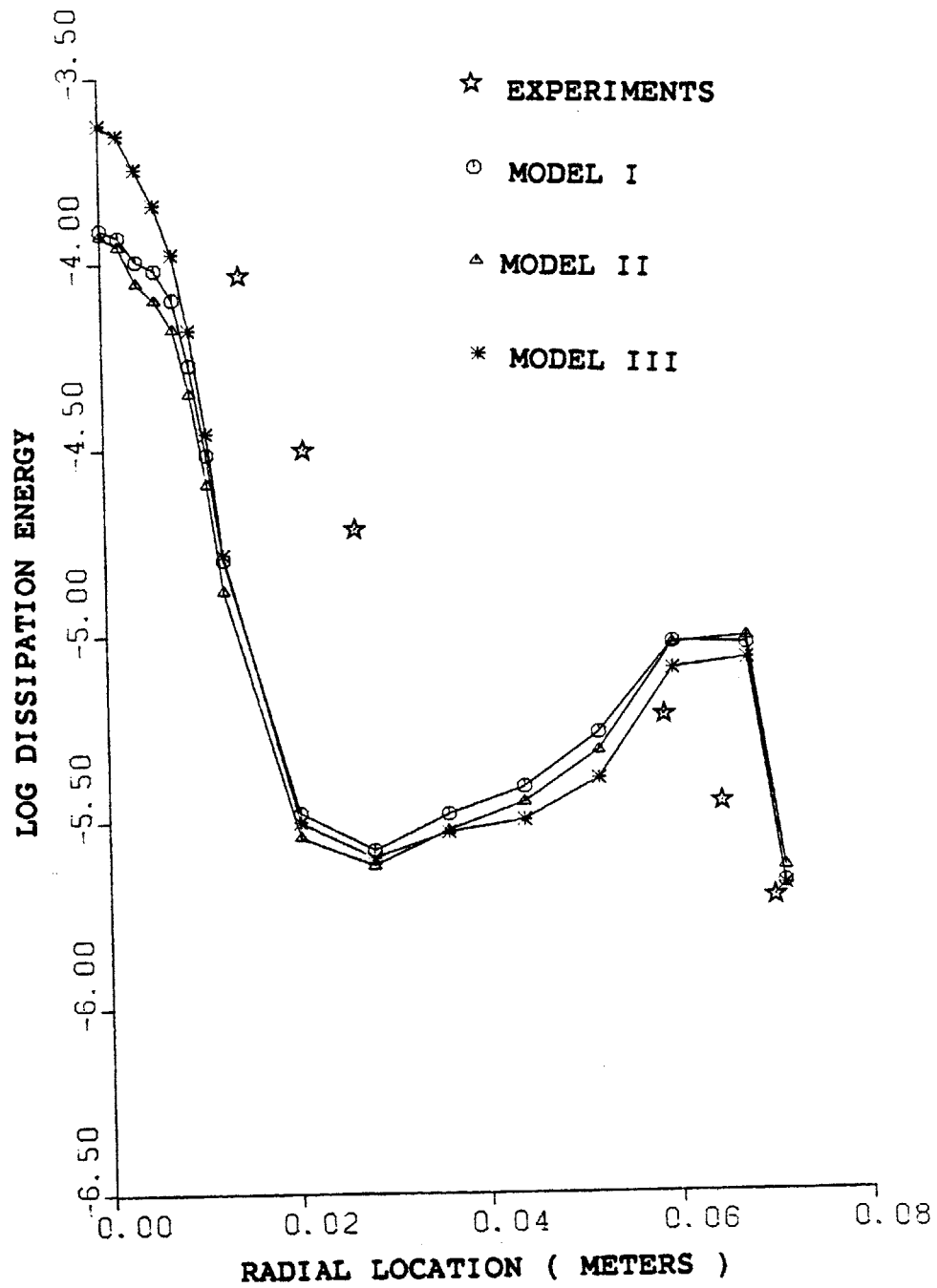


Fig. 5.15 Turbulence Dissipation Energy (W/Kg.s) Depth = 2.50 cms

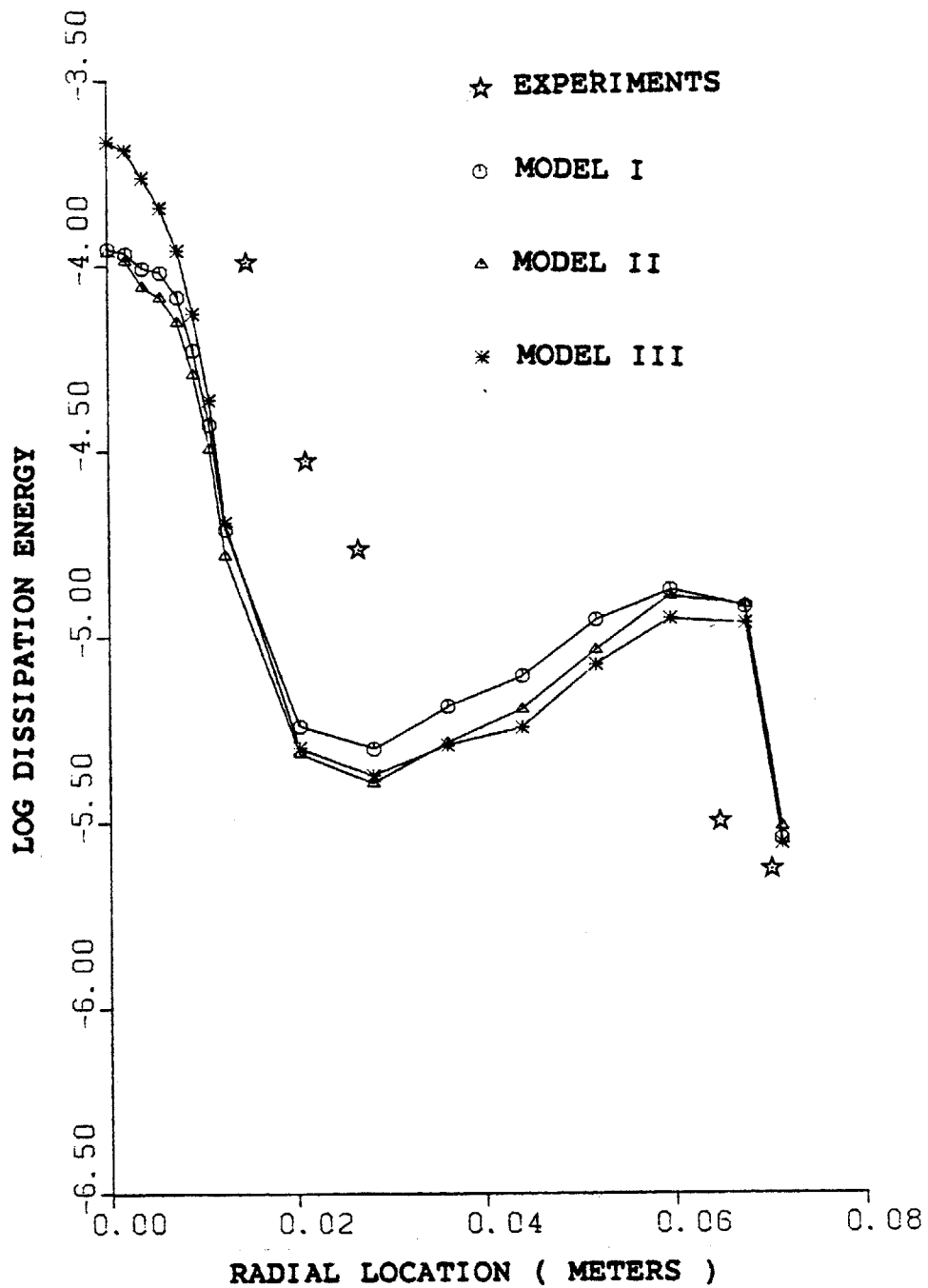


Fig. 5.16 Turbulence Dissipation Energy (W/Kg.s) Depth = 2.85 cms

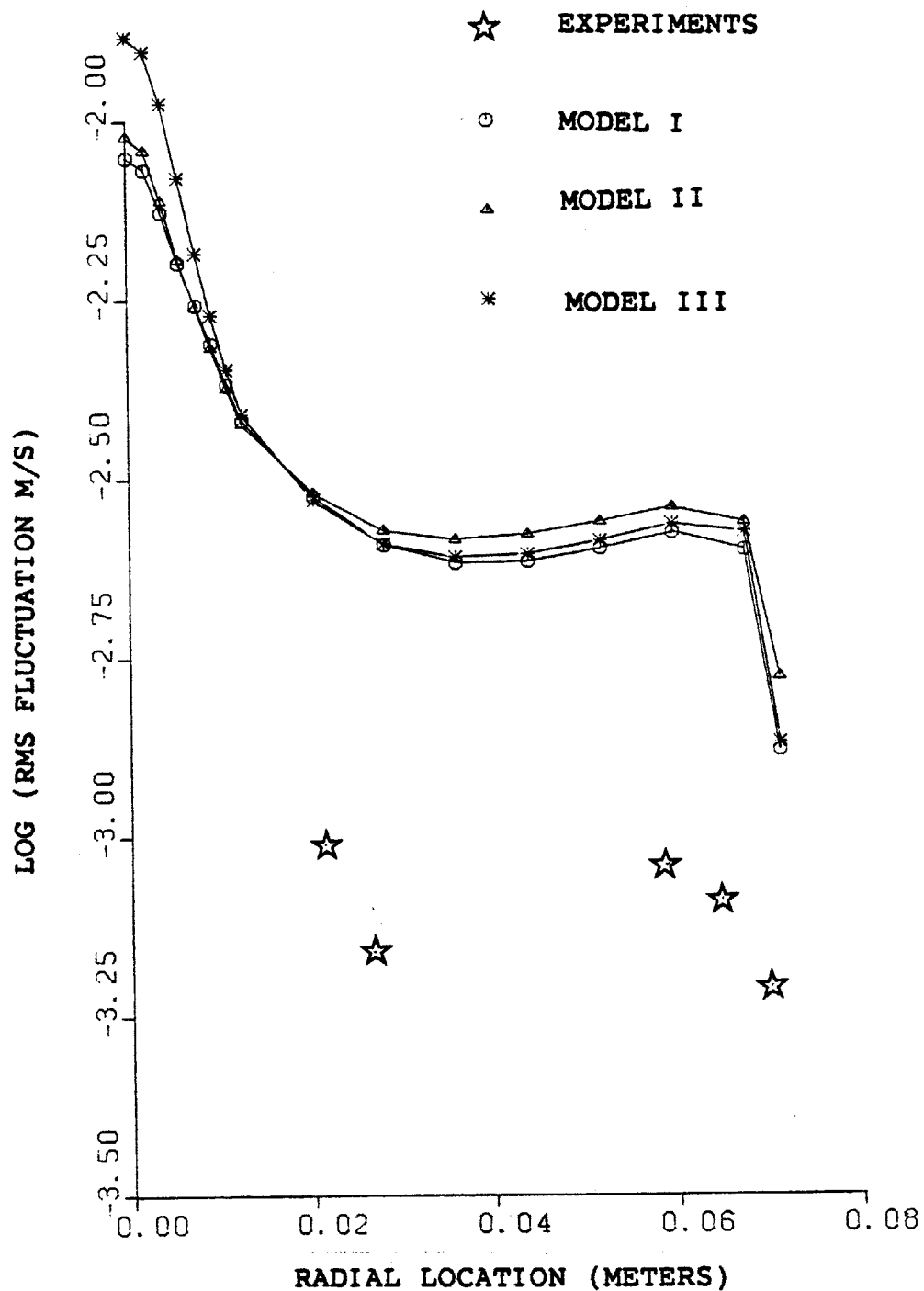


Fig. 5.17 R.M.S. Turbulent Velocity (M/sec) Depth = 0.70 cms

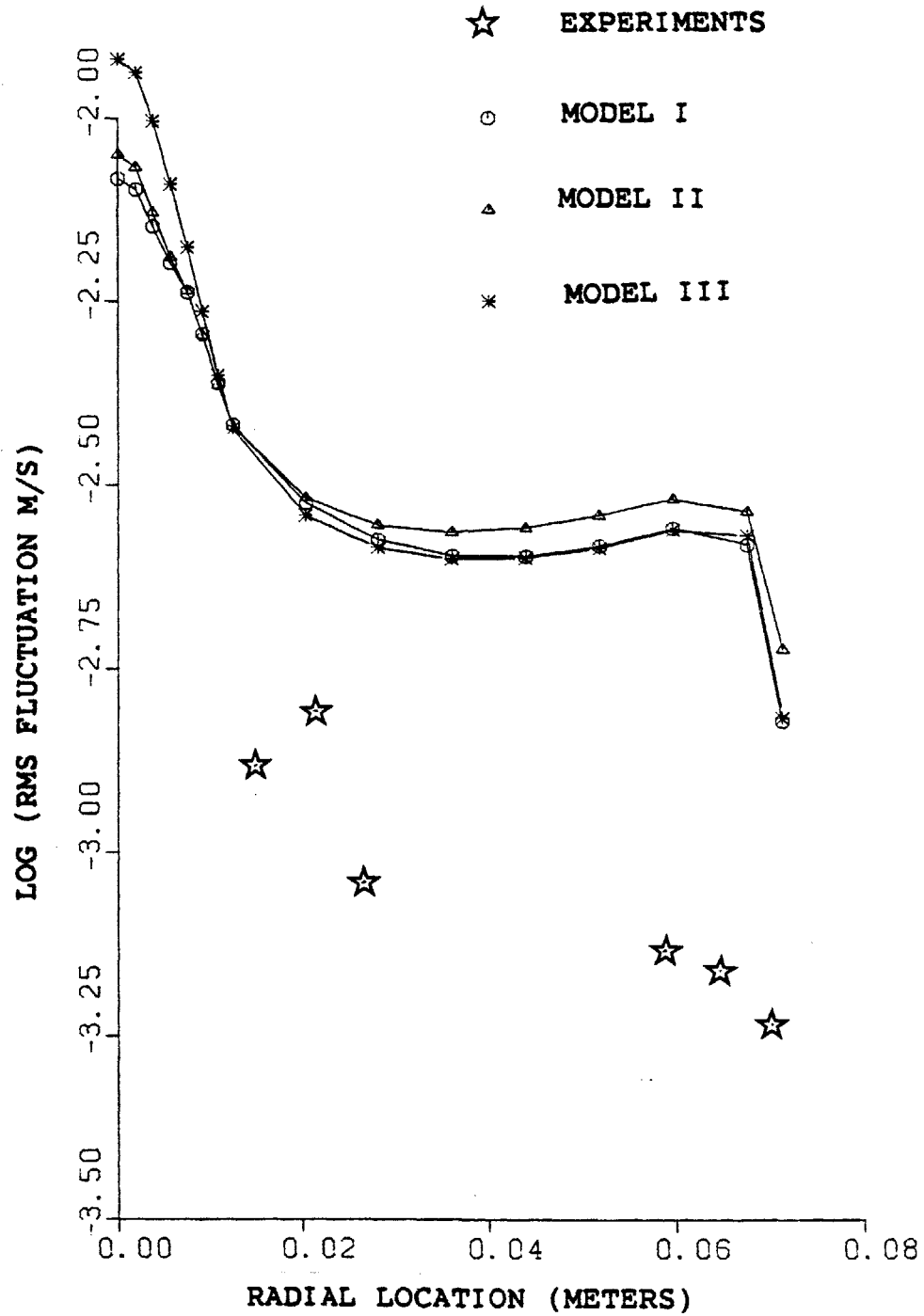


Fig. 5.18 R.M.S. Turbulent Velocity (M/sec) Depth = 1.05 cms

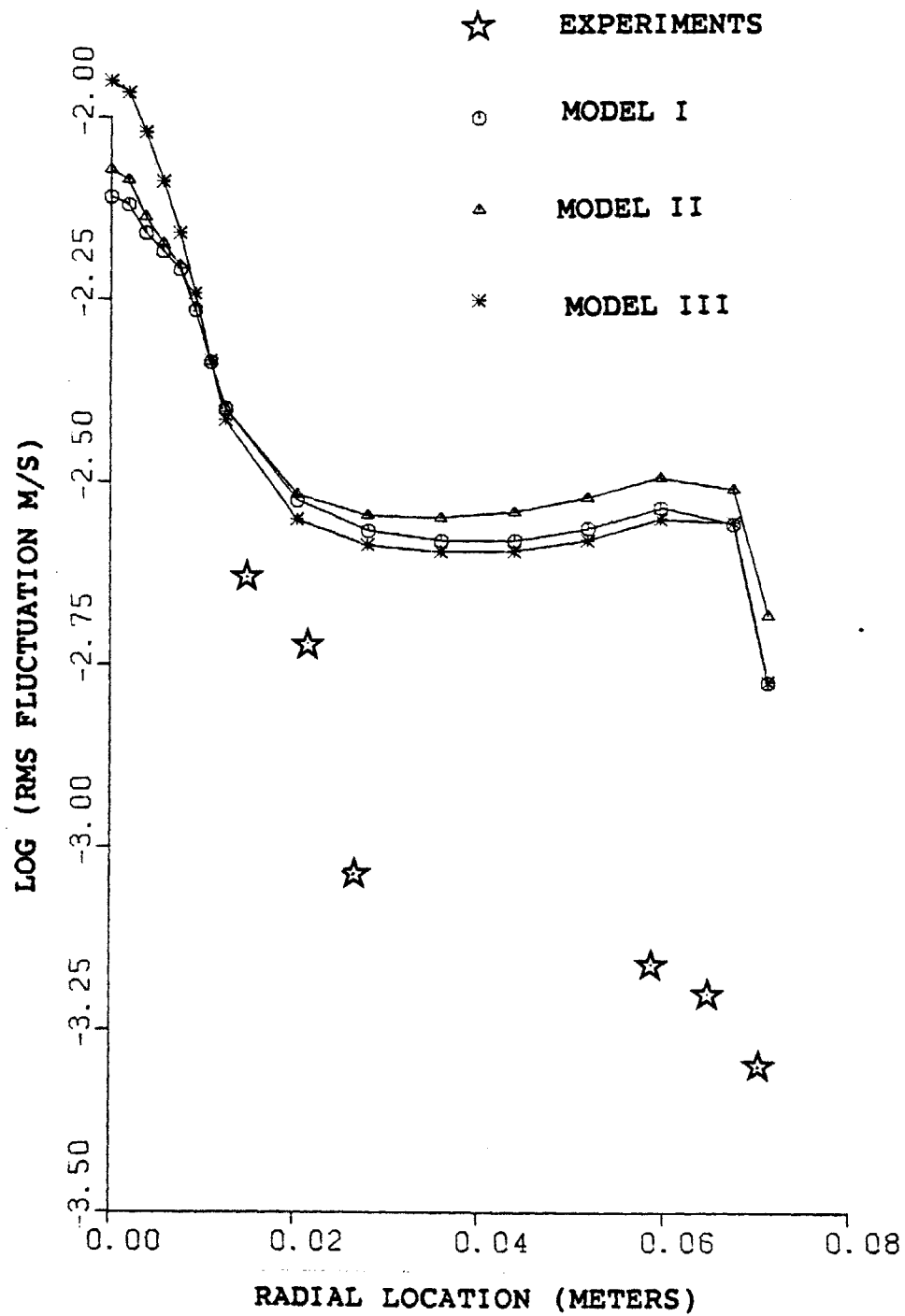


Fig. 5.19 R.M.S. Turbulent Velocity (M/sec) Depth = 1.44 cms

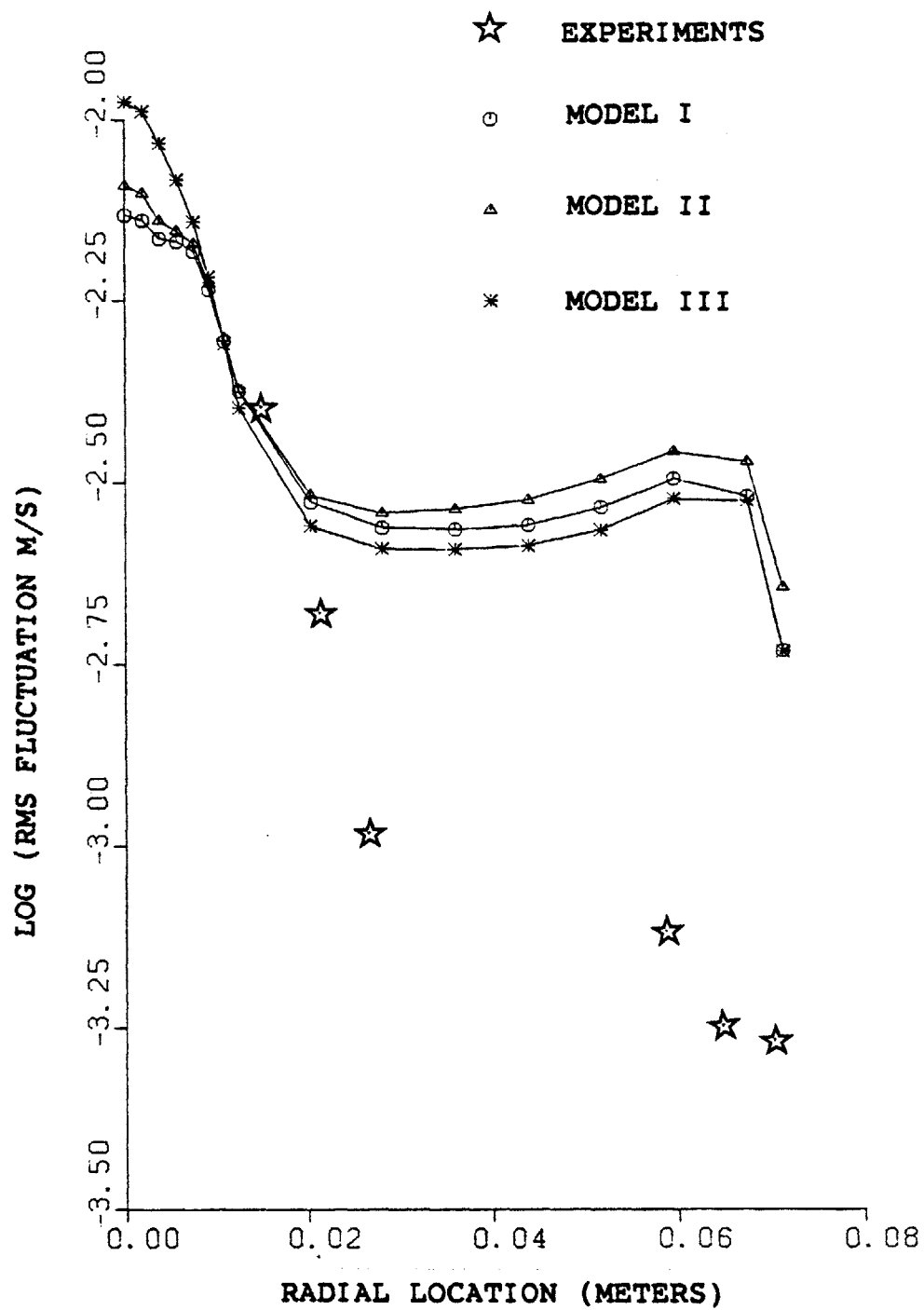


Fig. 5.20 R.M.S. Turbulent Velocity (M/sec) Depth = 1.79 cms



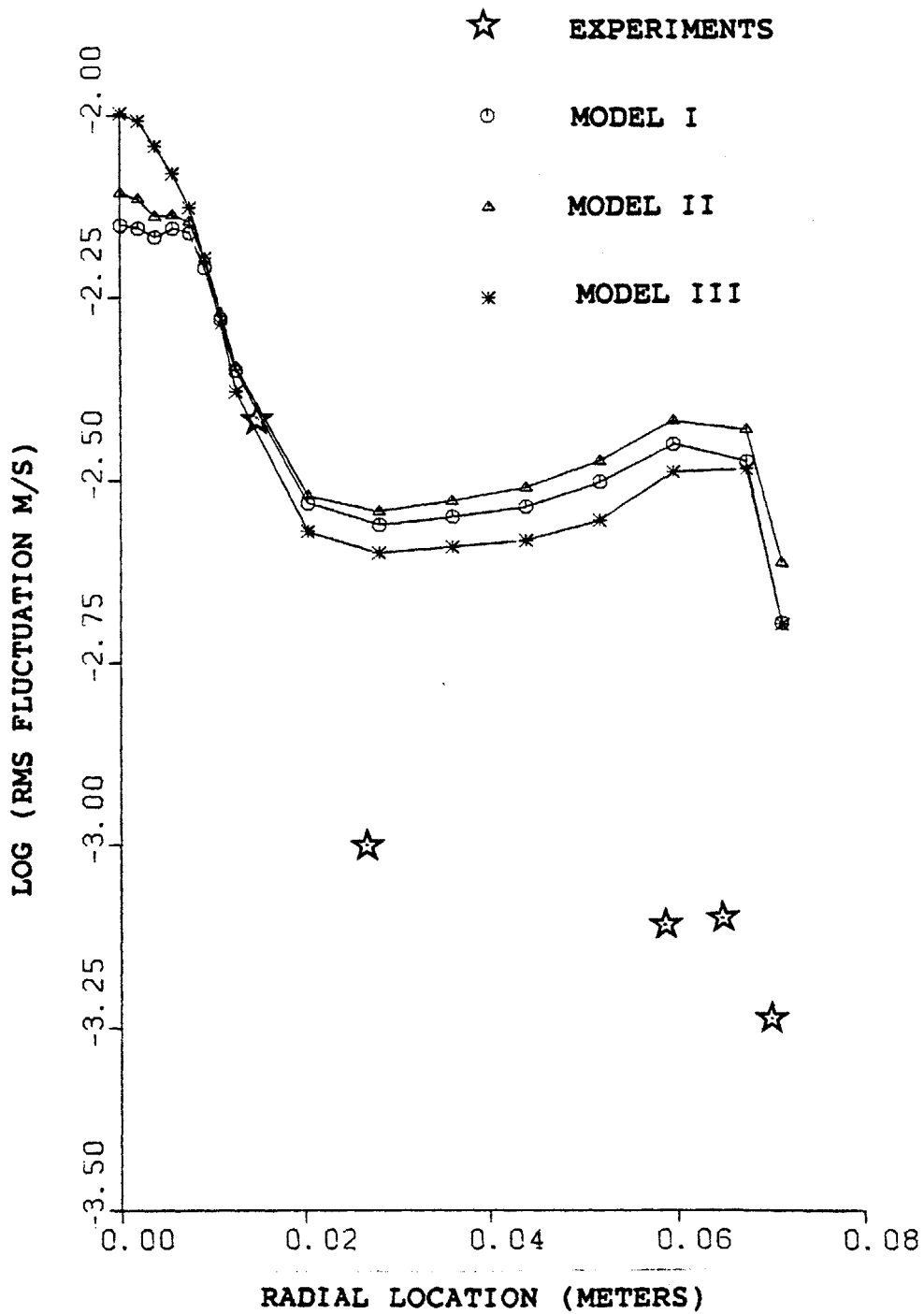


Fig. 5.21 R.M.S. Turbulent Velocity (M/sec) Depth = 2.14 cms

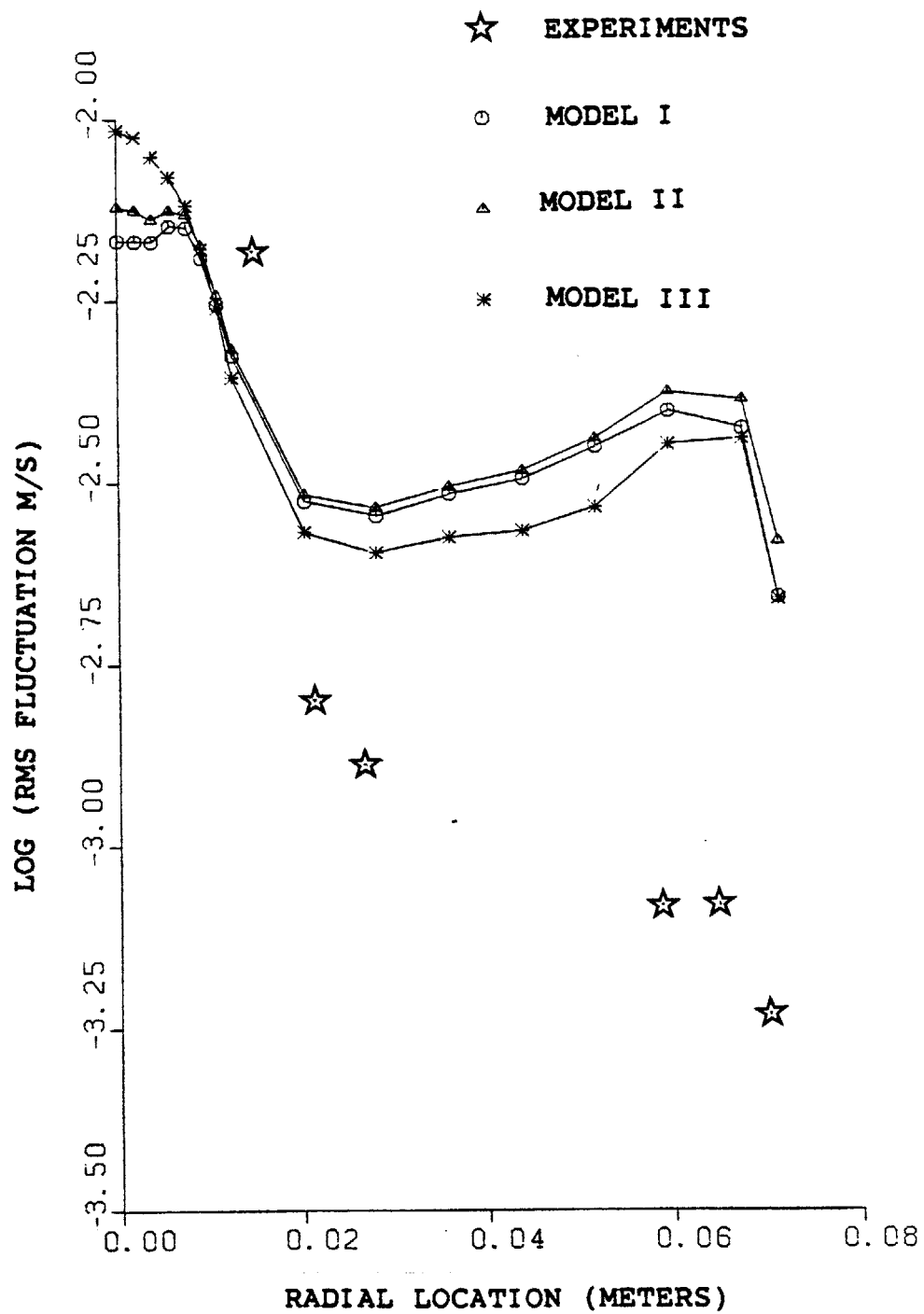


Fig. 5.22 R.M.S. Turbulent Velocity (M/sec) Depth = 2.50 cms

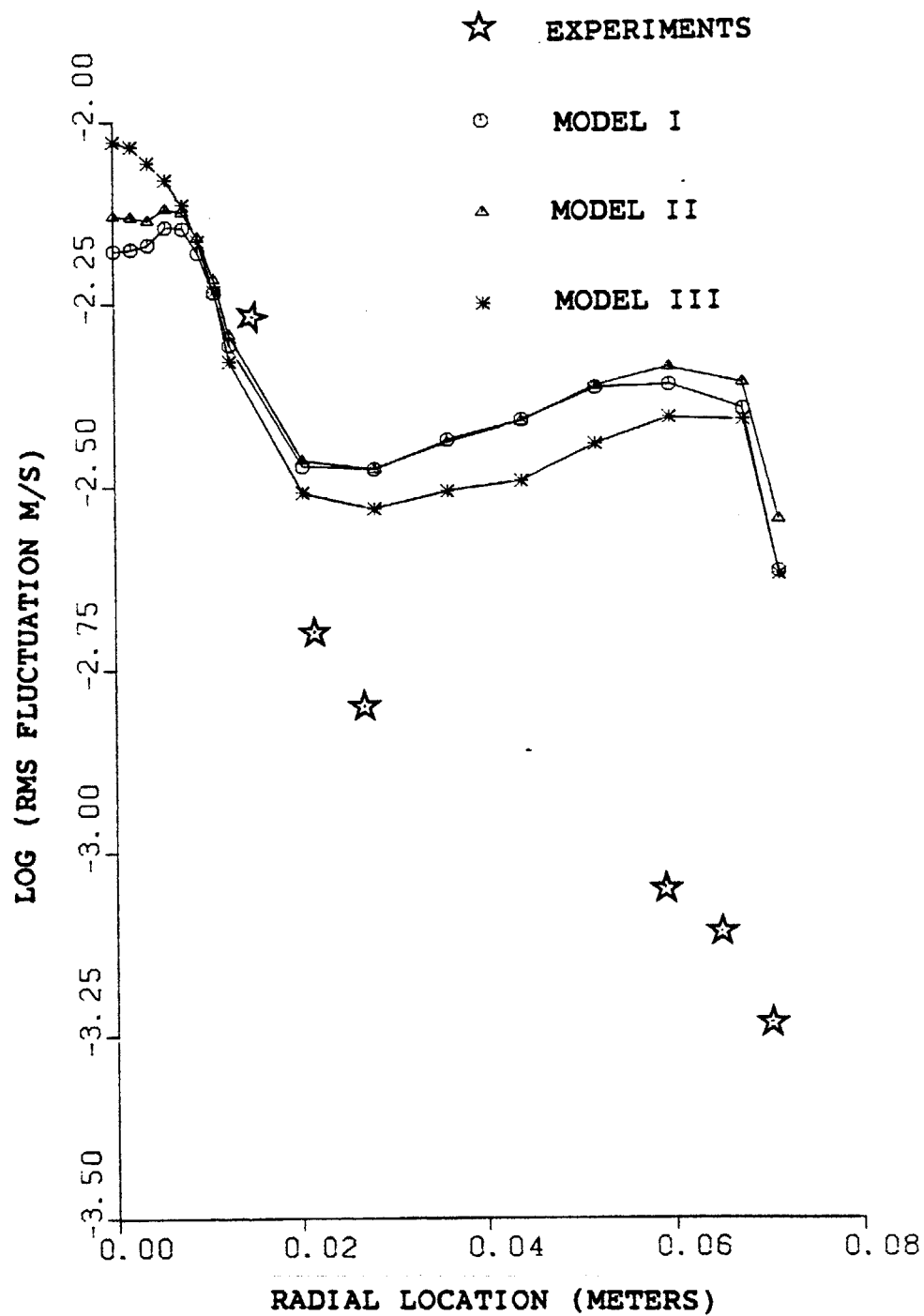


Fig. 5.23 R.M.S. Turbulent Velocity (M/sec) Depth = 2.85 cms

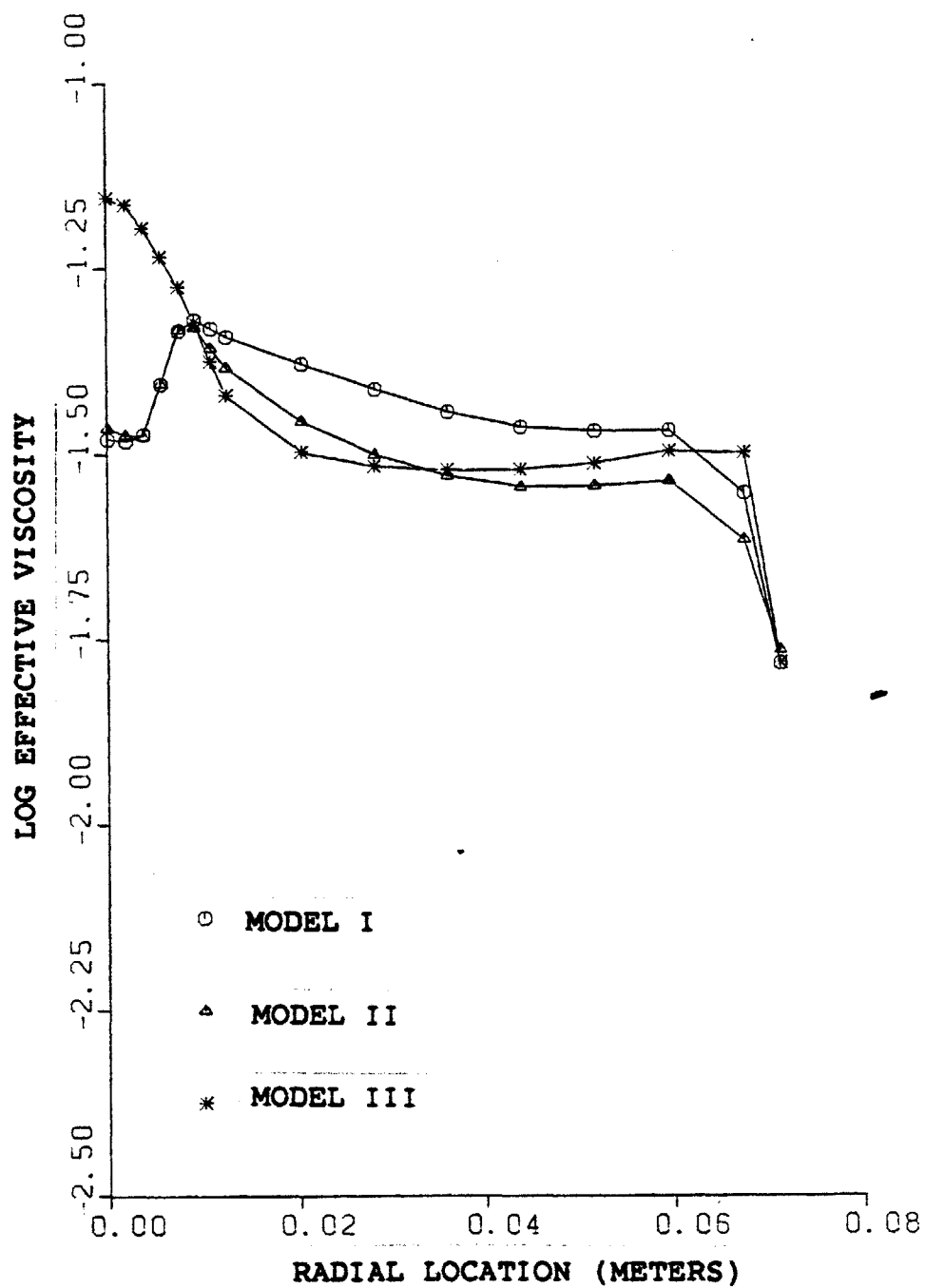


Fig. 5.24 Effective Viscosity (Kg/m-sec) Depth = 1.44 cms

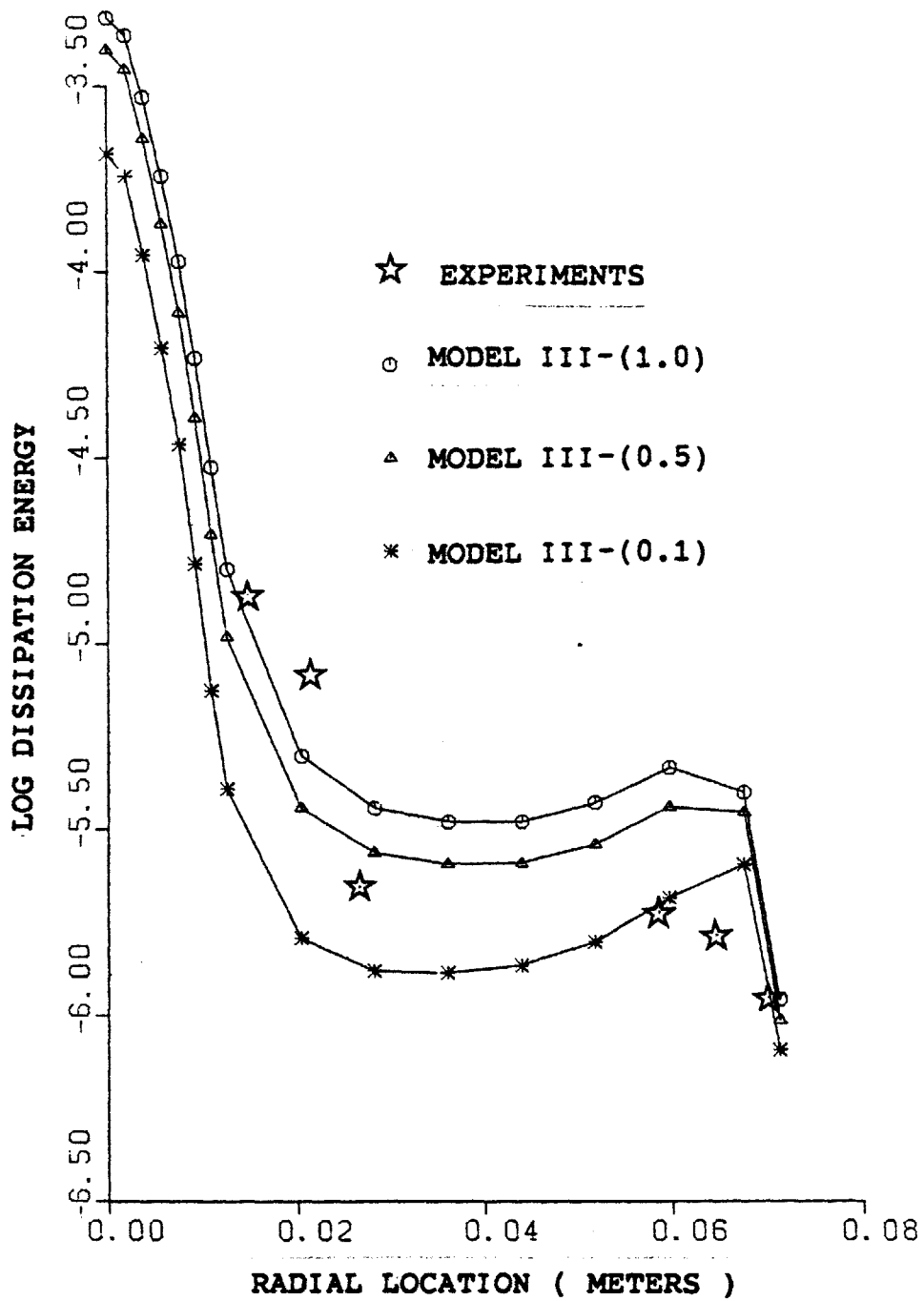


Fig. 5.25 Effect of Varying  $\eta$  on D.Energy Profile, Depth = 1.44 cms

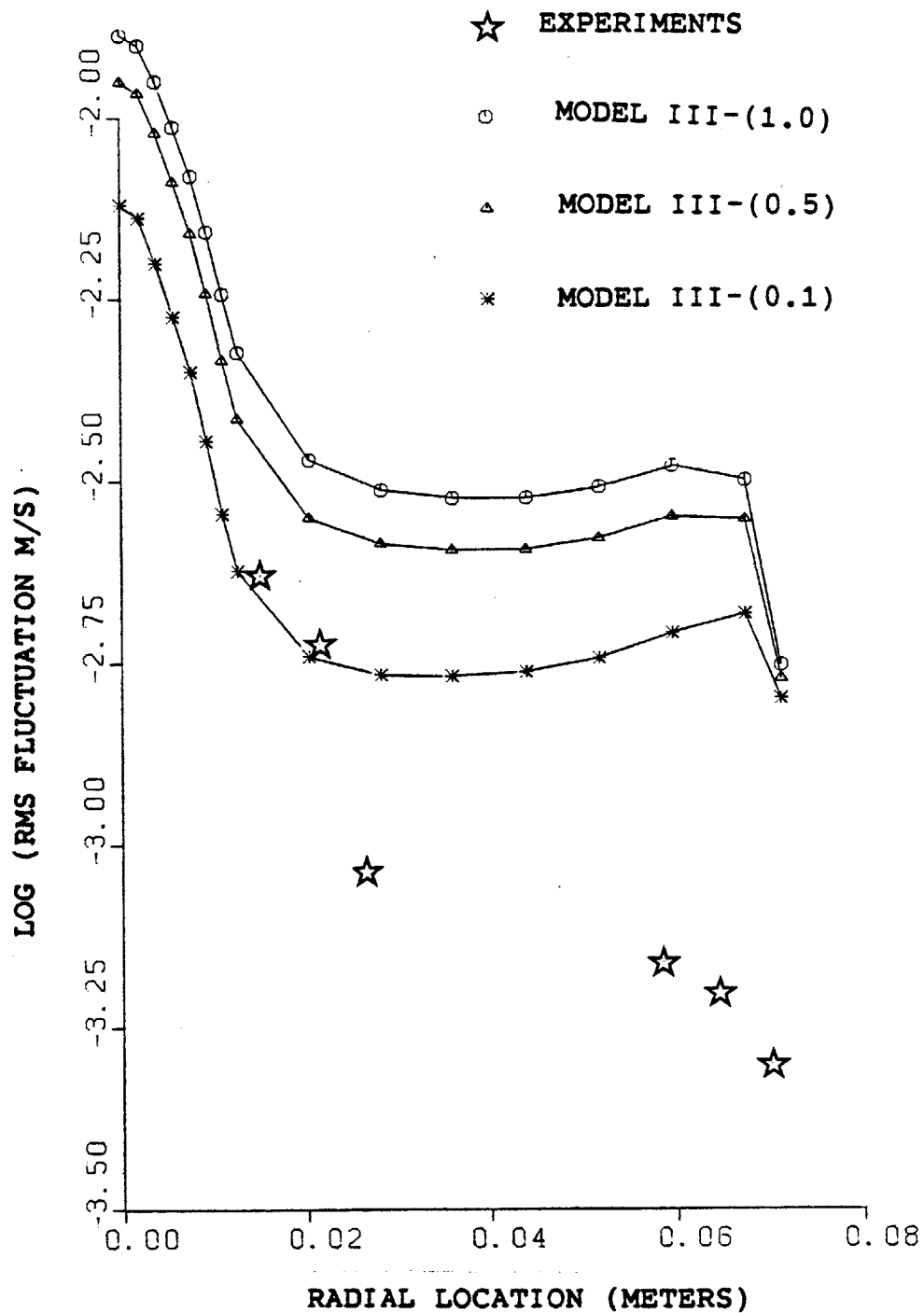


Fig. 5.26 Effect of Varying  $\eta$  on Fluc. Velocity, Depth = 1.44 cms

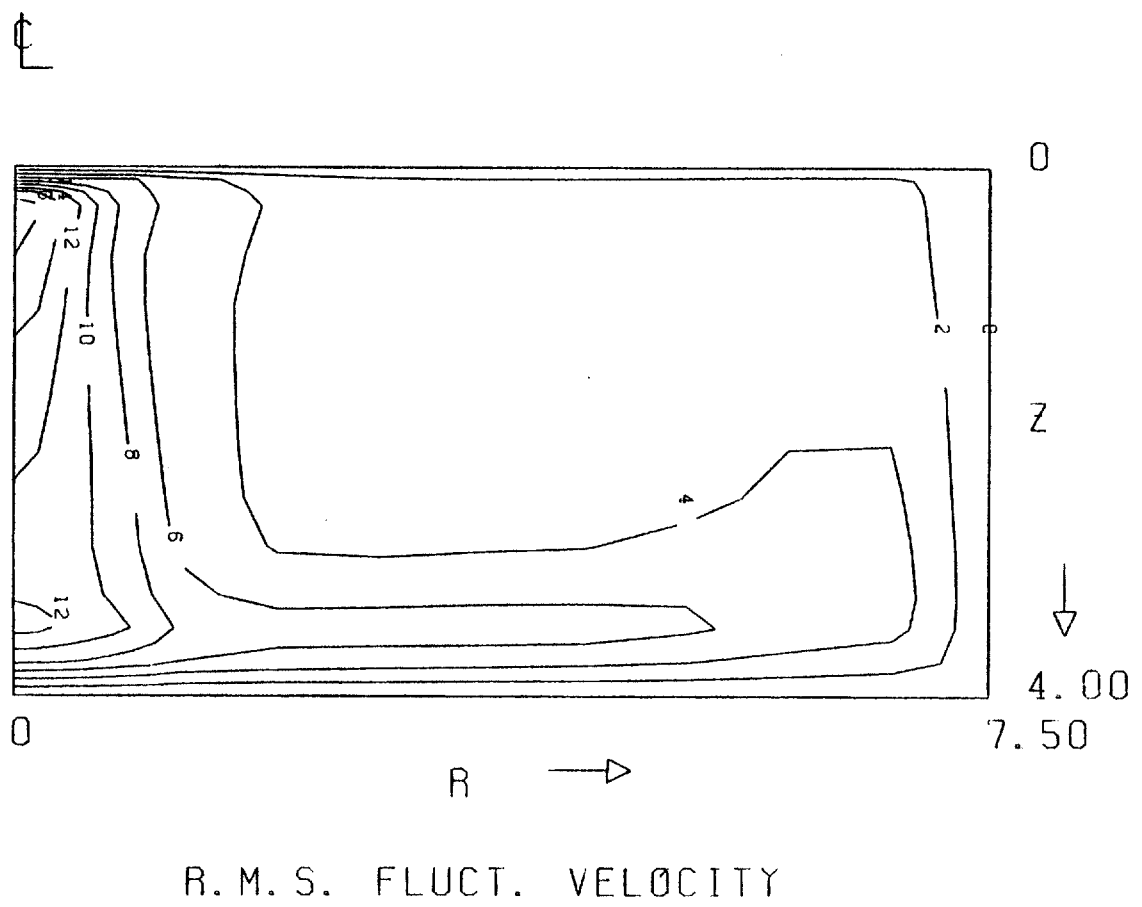
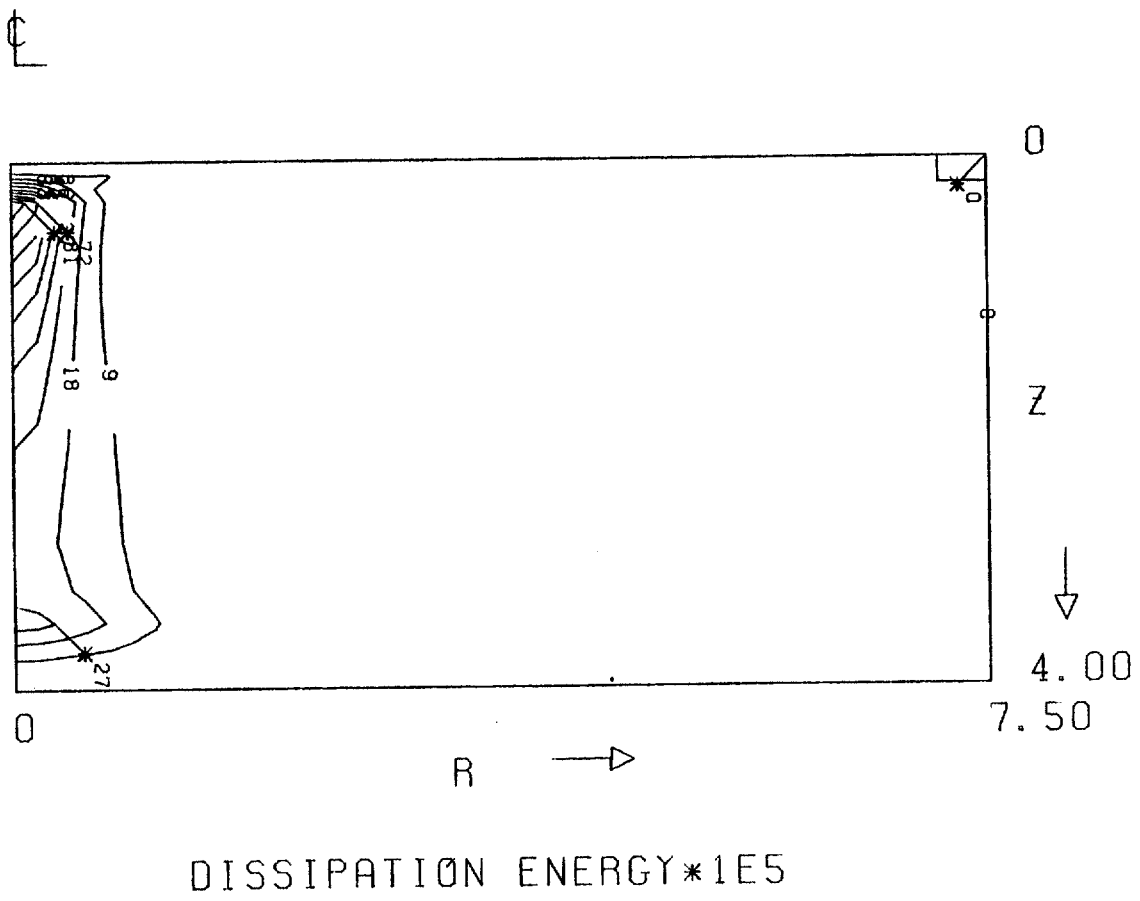


Fig. 5.27 Contour Plot of Fluc. Velocity distribution in Melt

Fig. 5.28 Contour Plot of D. Energy Distribution in Melt





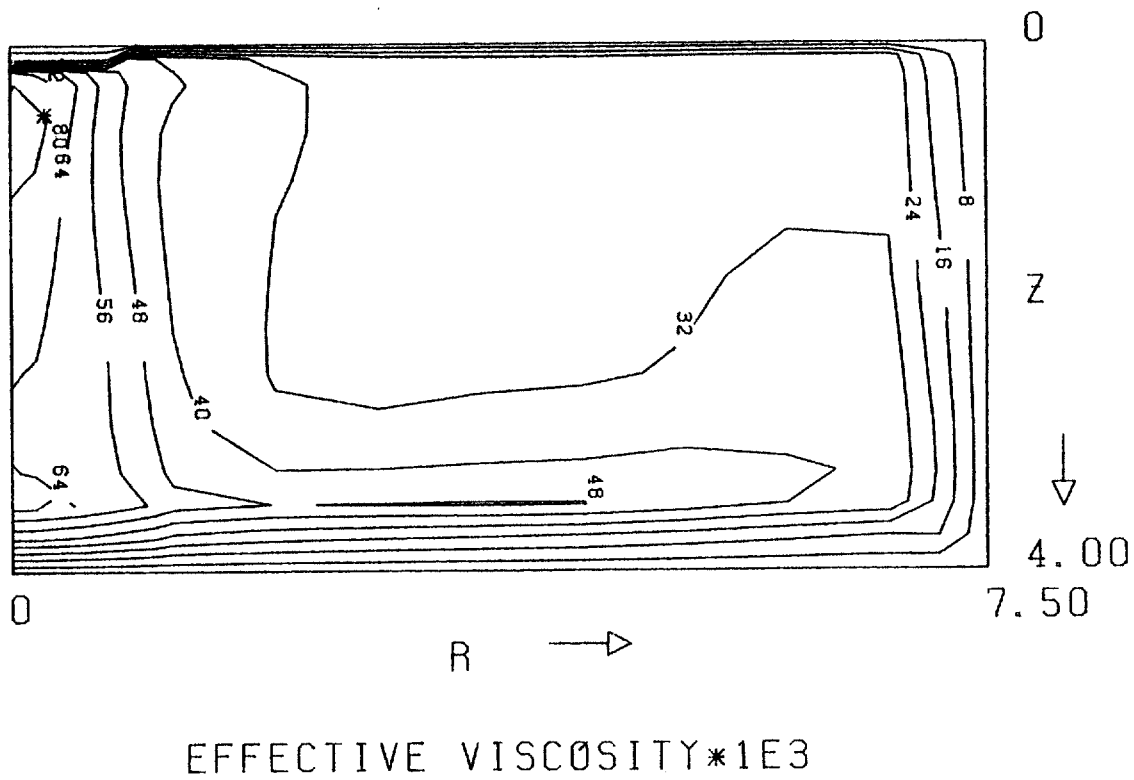
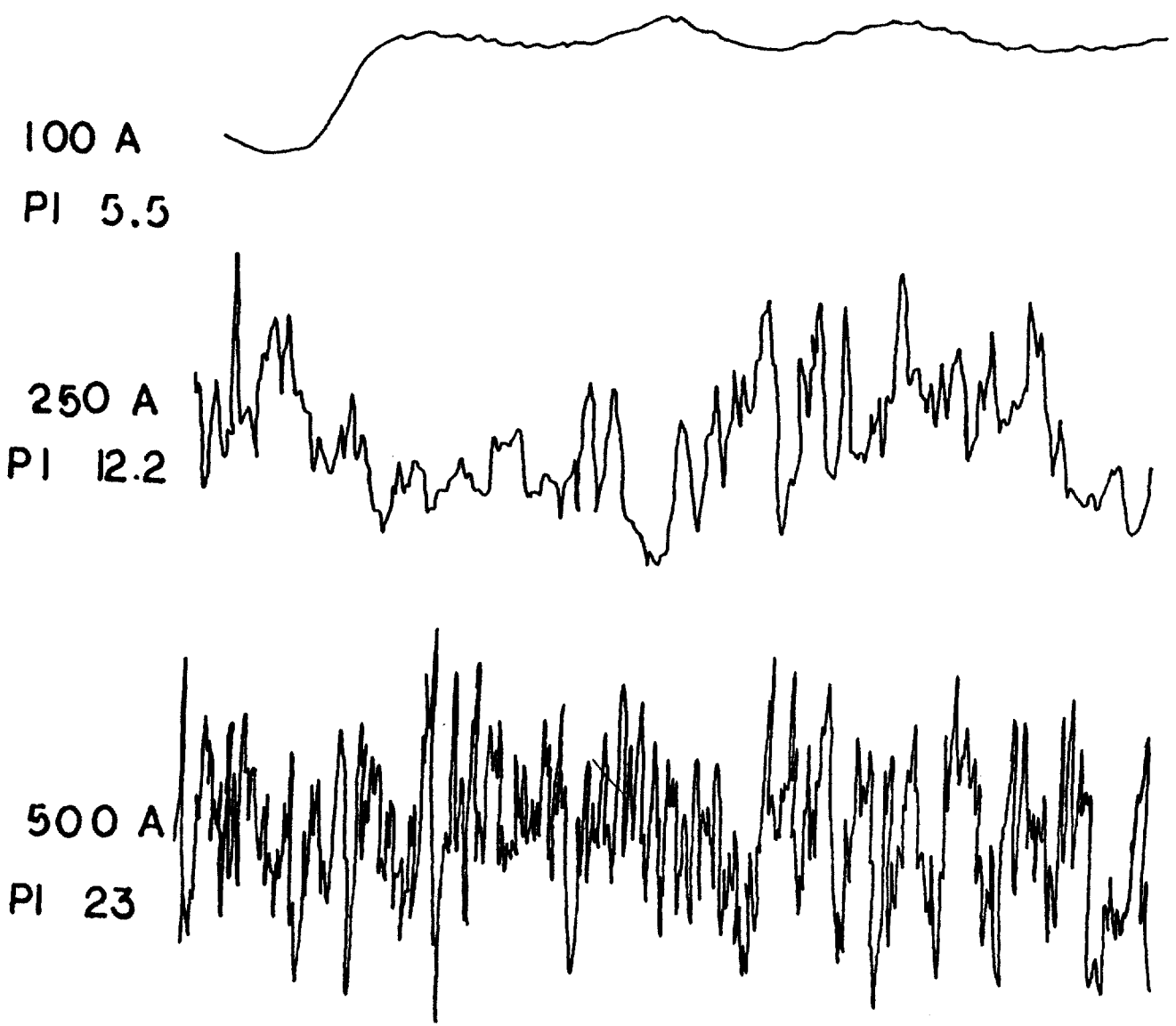


Fig. 5.29 Contour Plot of Effective Viscosity Distribution in Melt

Fig. 5.30 Typical Anemometer Voltage outputs with computed values of PI



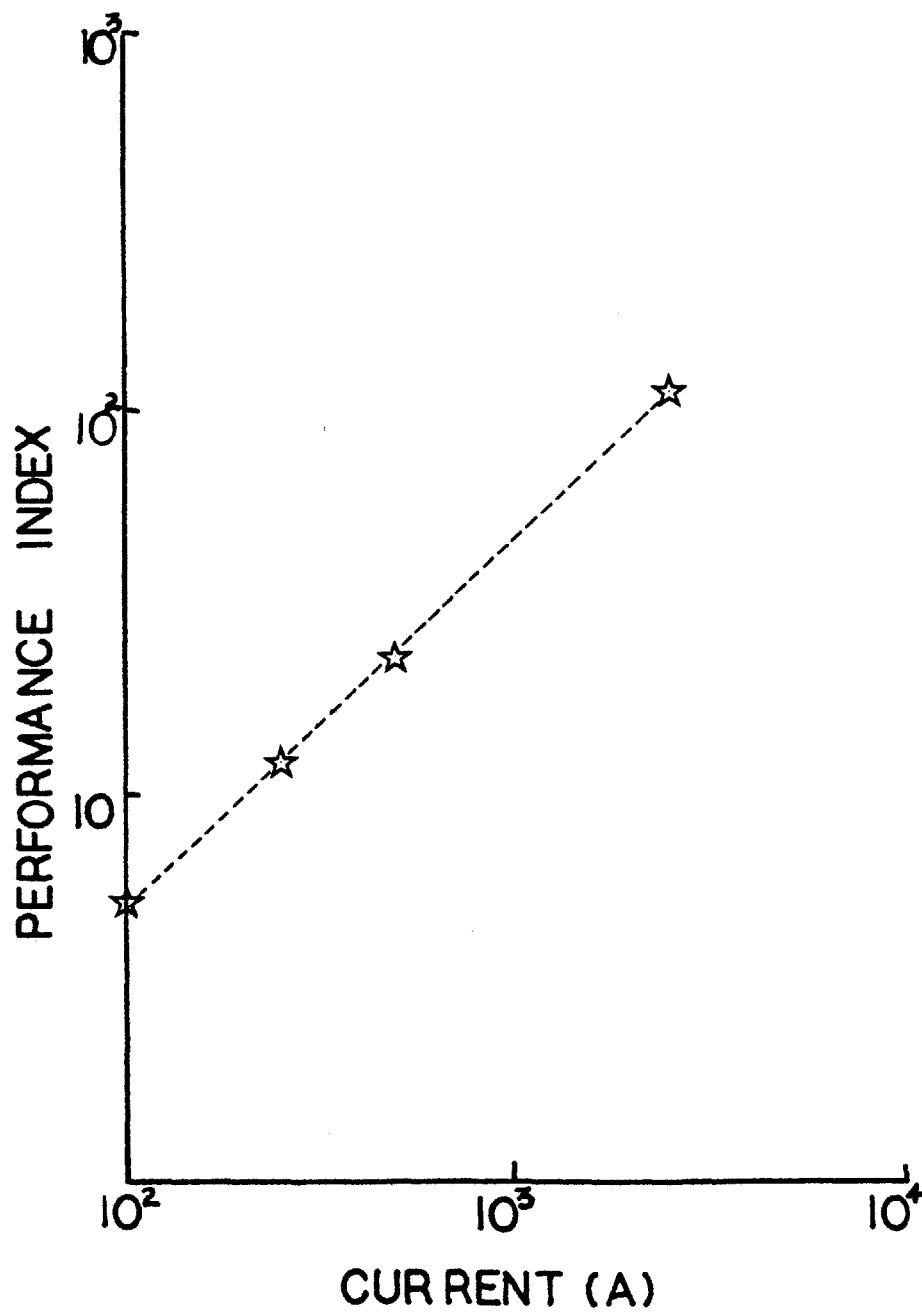


Fig. 5.31 Variation of Computed value of PI with current passed through melt

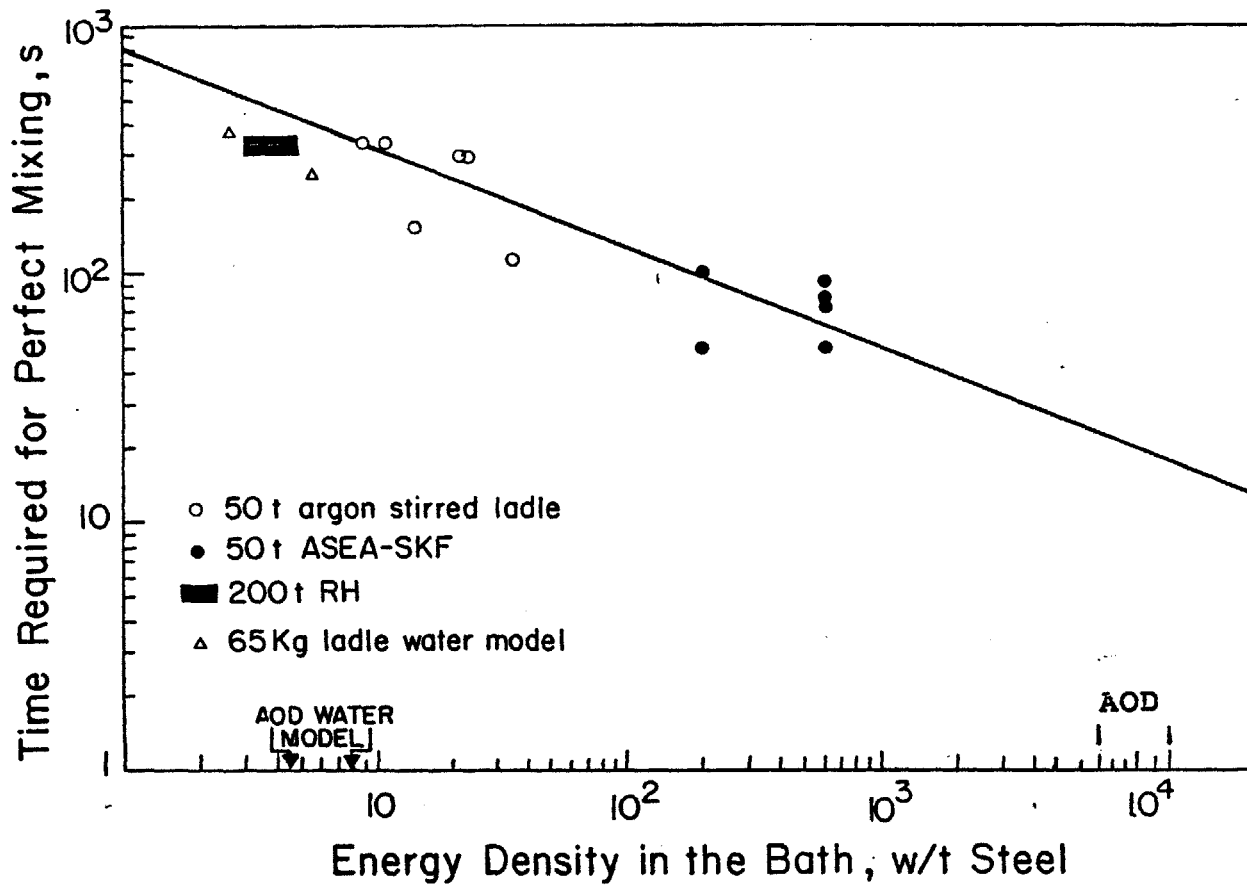


Fig. 5.32 Variation of Mixing Time with Stirring Power density in the melt (Ref. 59)

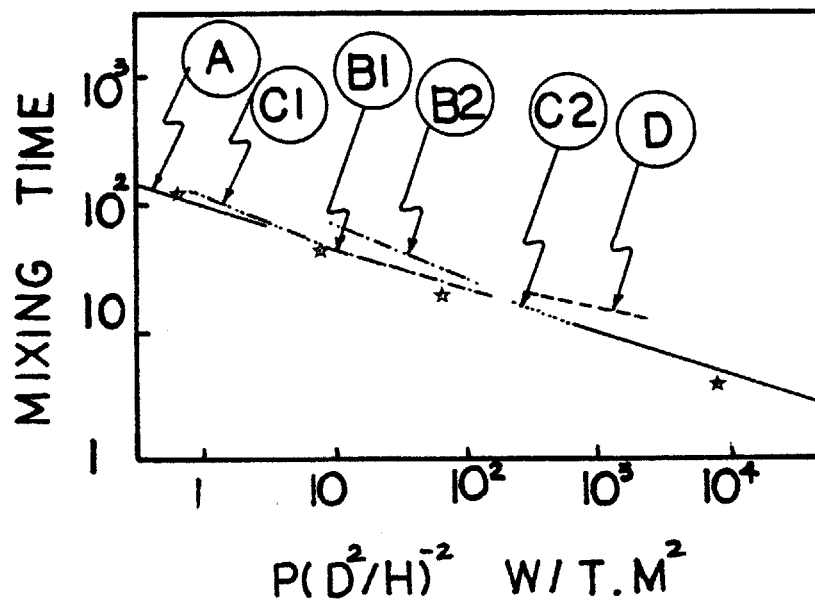


Fig. 5.33 Variation of Mixing Time with  $(P/D^2 / H)$  in the melt (Ref.58)

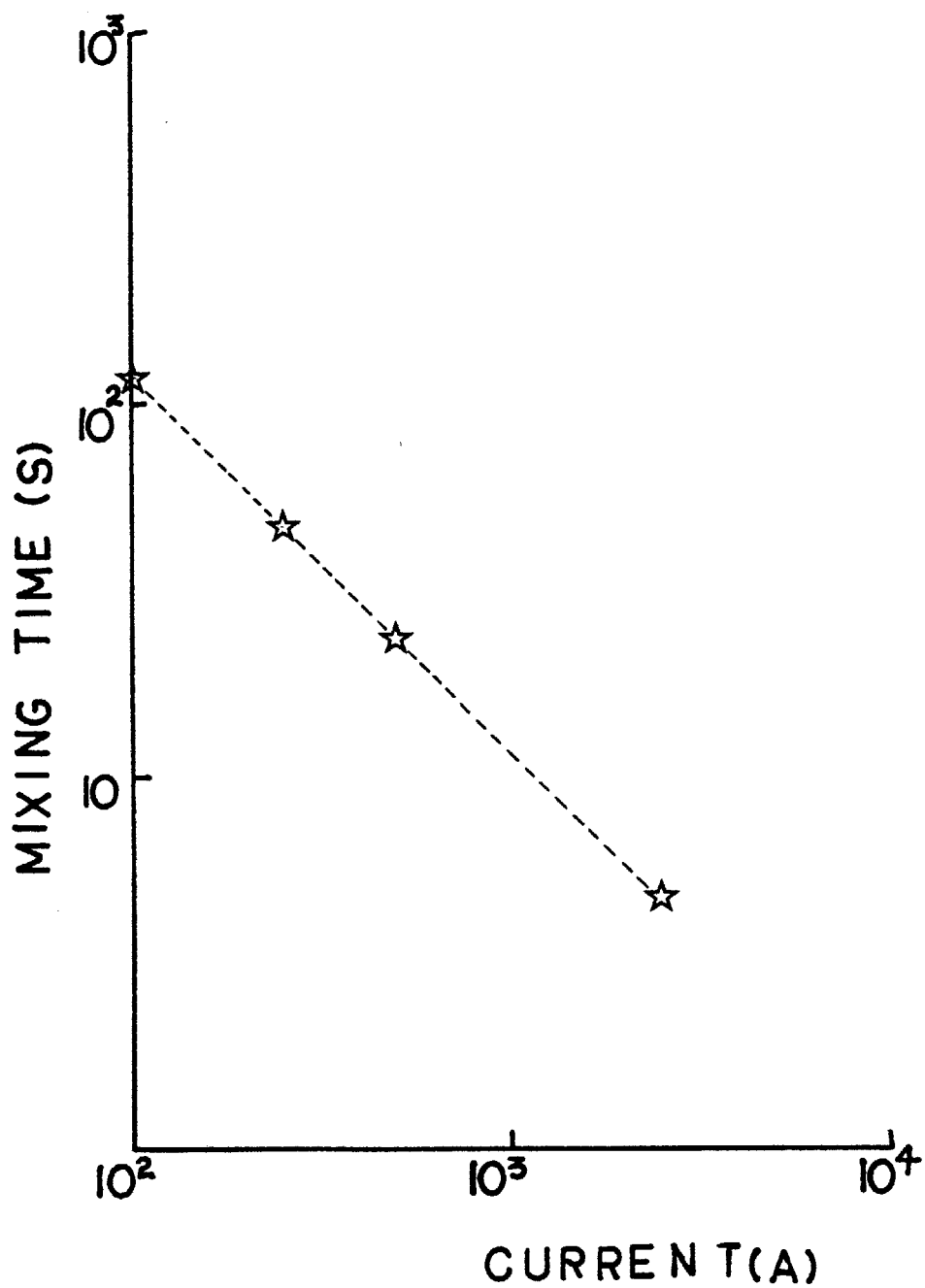
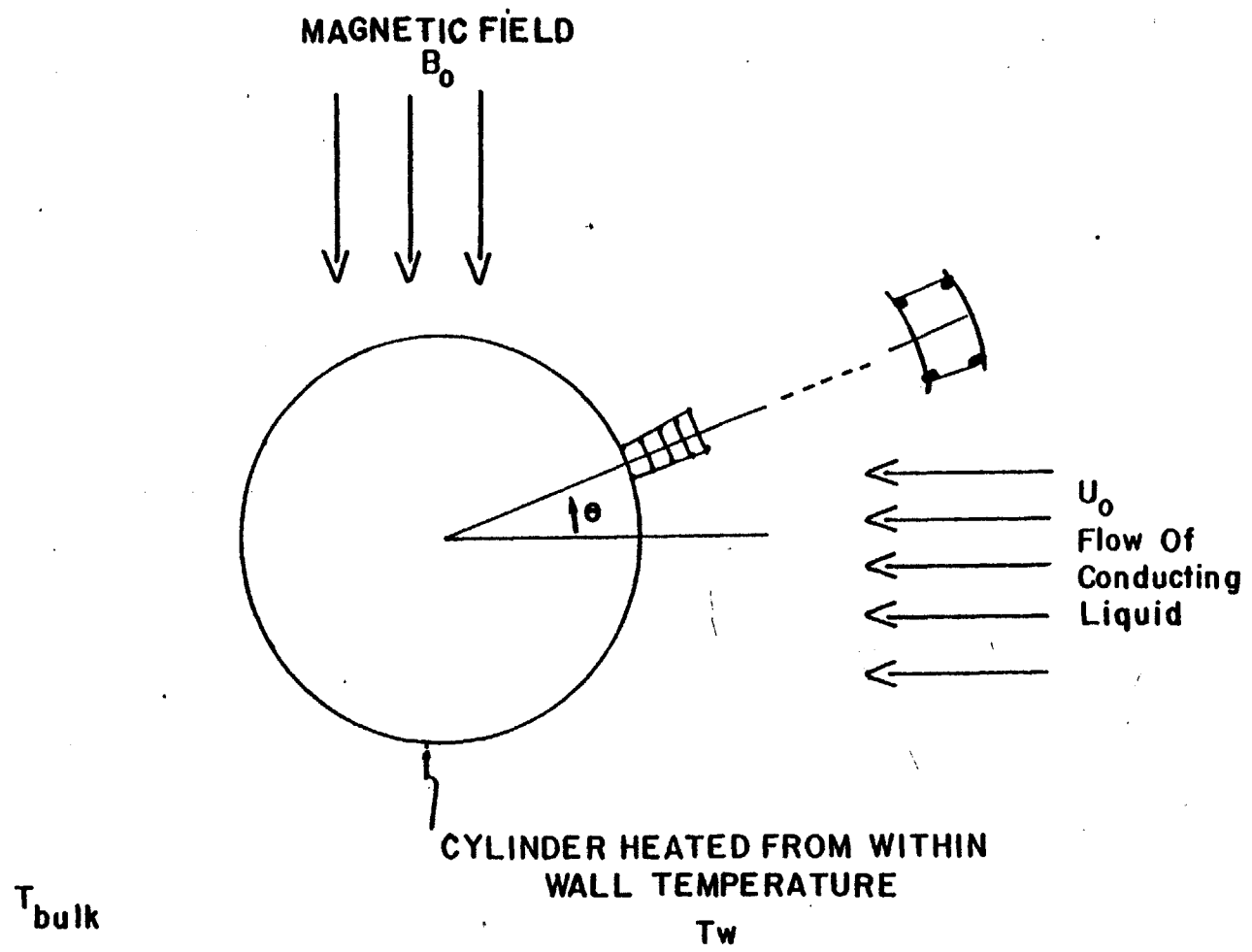


Fig. 5.34 Computed Variation of Mixing Time with Current passed through melt



Fig. 5.36 Flow Configuration for MHD flow around cylinder





- Comparison of calculated Drag  
Coefficients at Red = 40

Source	$C_D$
Kawaguti	1.618
Apelt	1.469
Kalis, Tsinober et al.	1.645
This study	1.616

Fig. 5.37 Comparison of Drag Computed Drag Coefficient at  
Mag.Field = 0

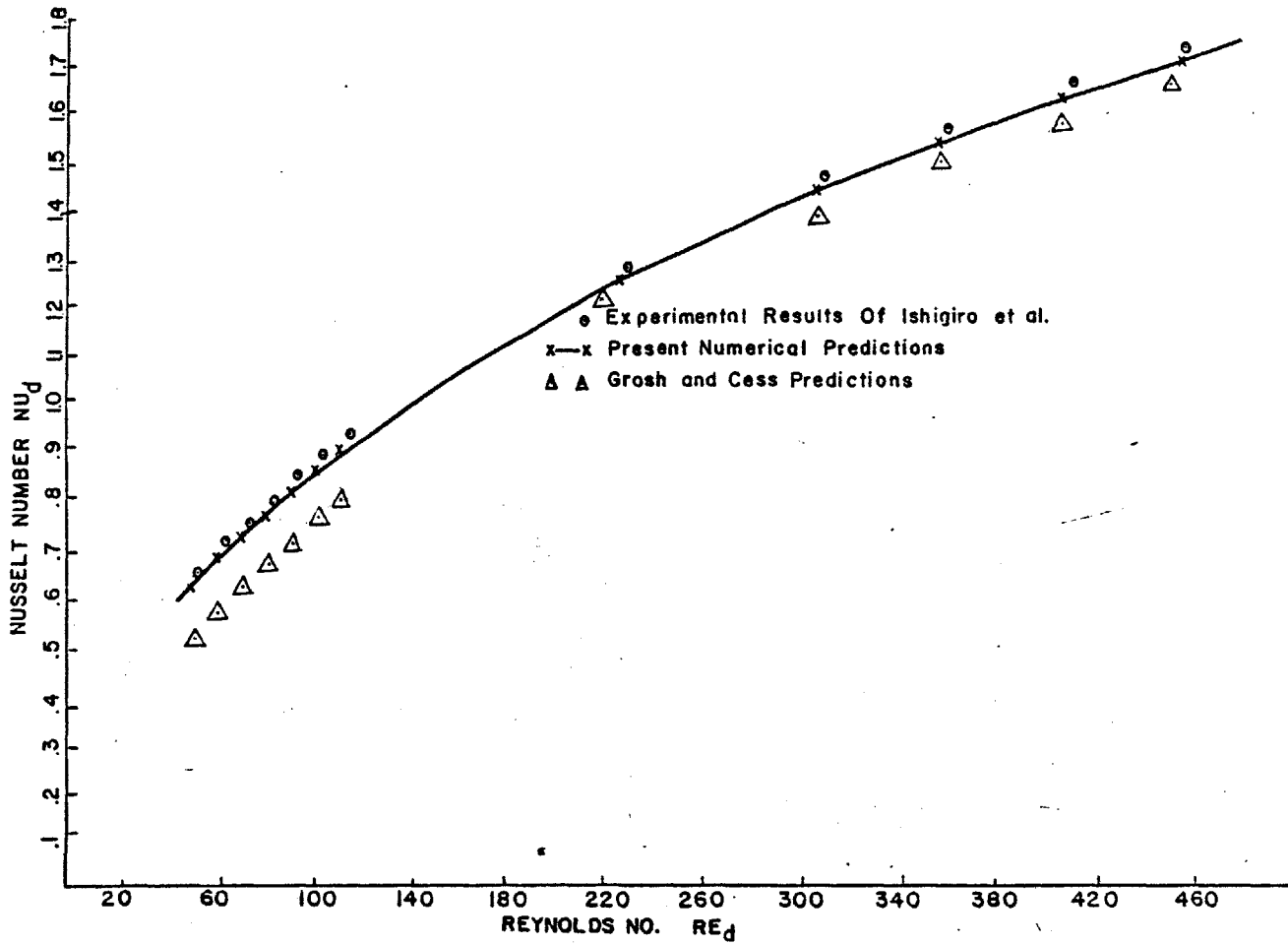


Fig. 5.38 Comparison of Averaged Nusselt Number with experiments

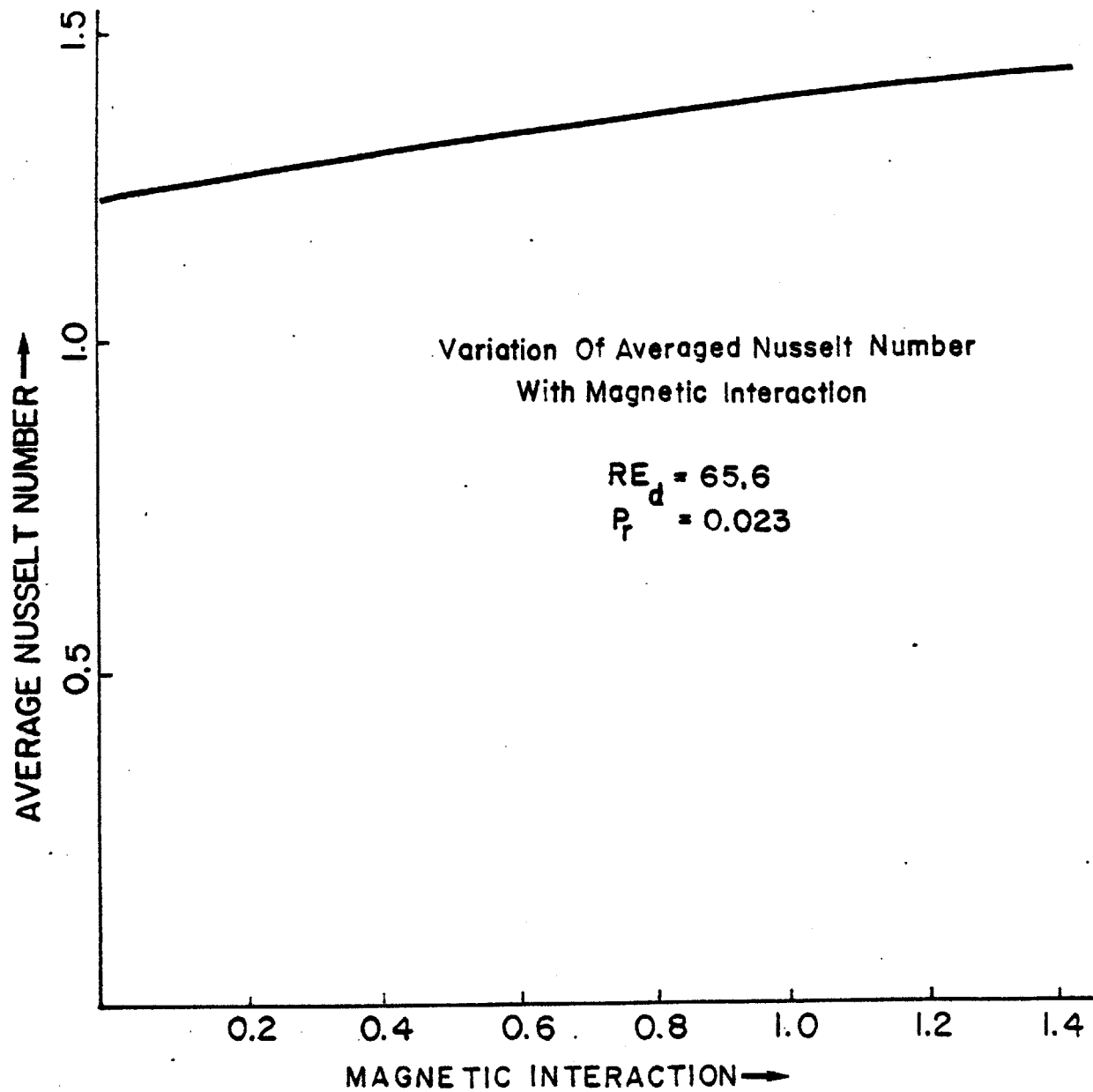


Fig. 5.39 Computed Variation of Averaged Nusselt Number with Magnetic Interaction Parameter

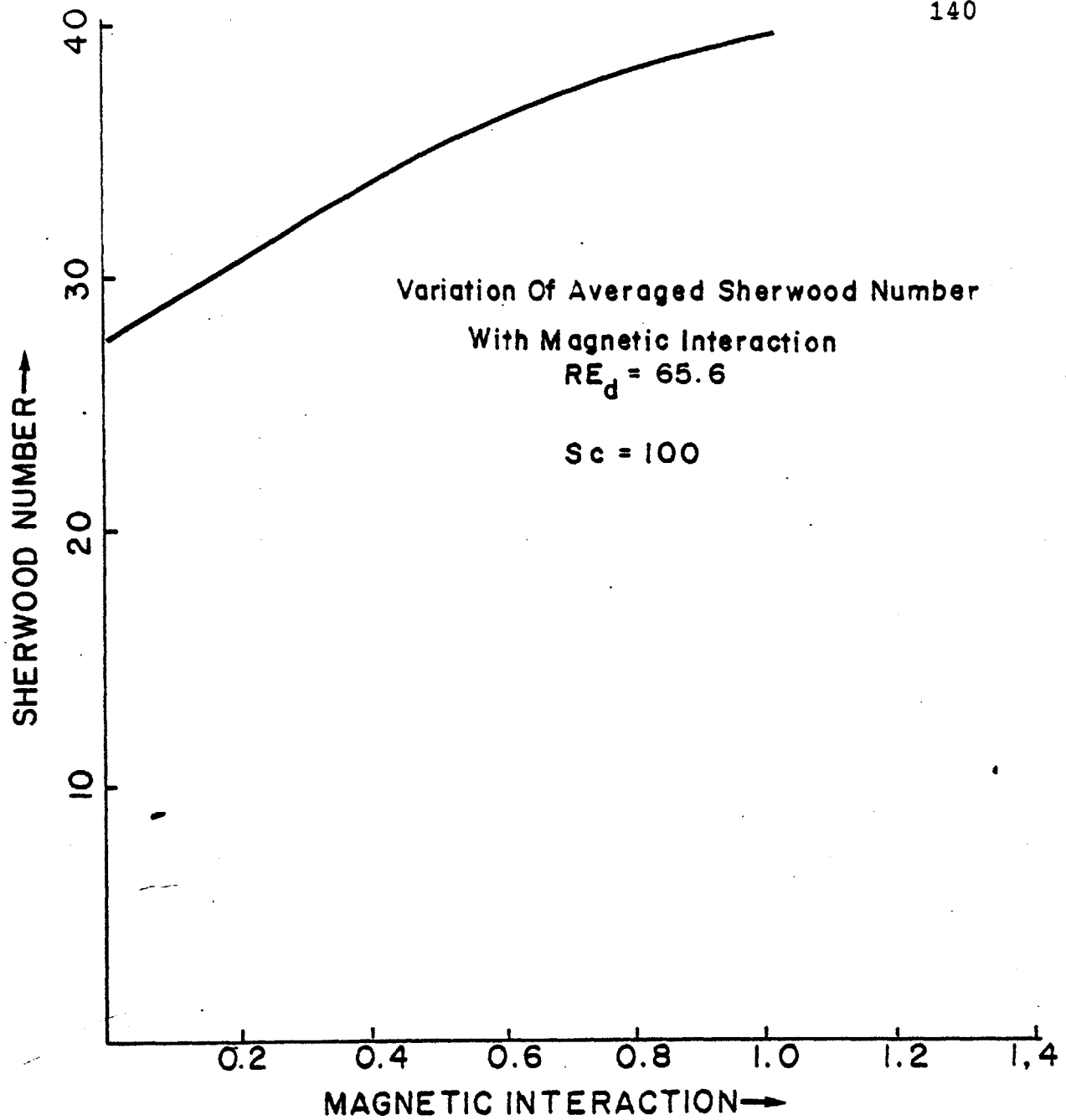


Fig. 5.40 Computed Variation of Averaged Sherwood Number with Magnetic Interaction Parameter

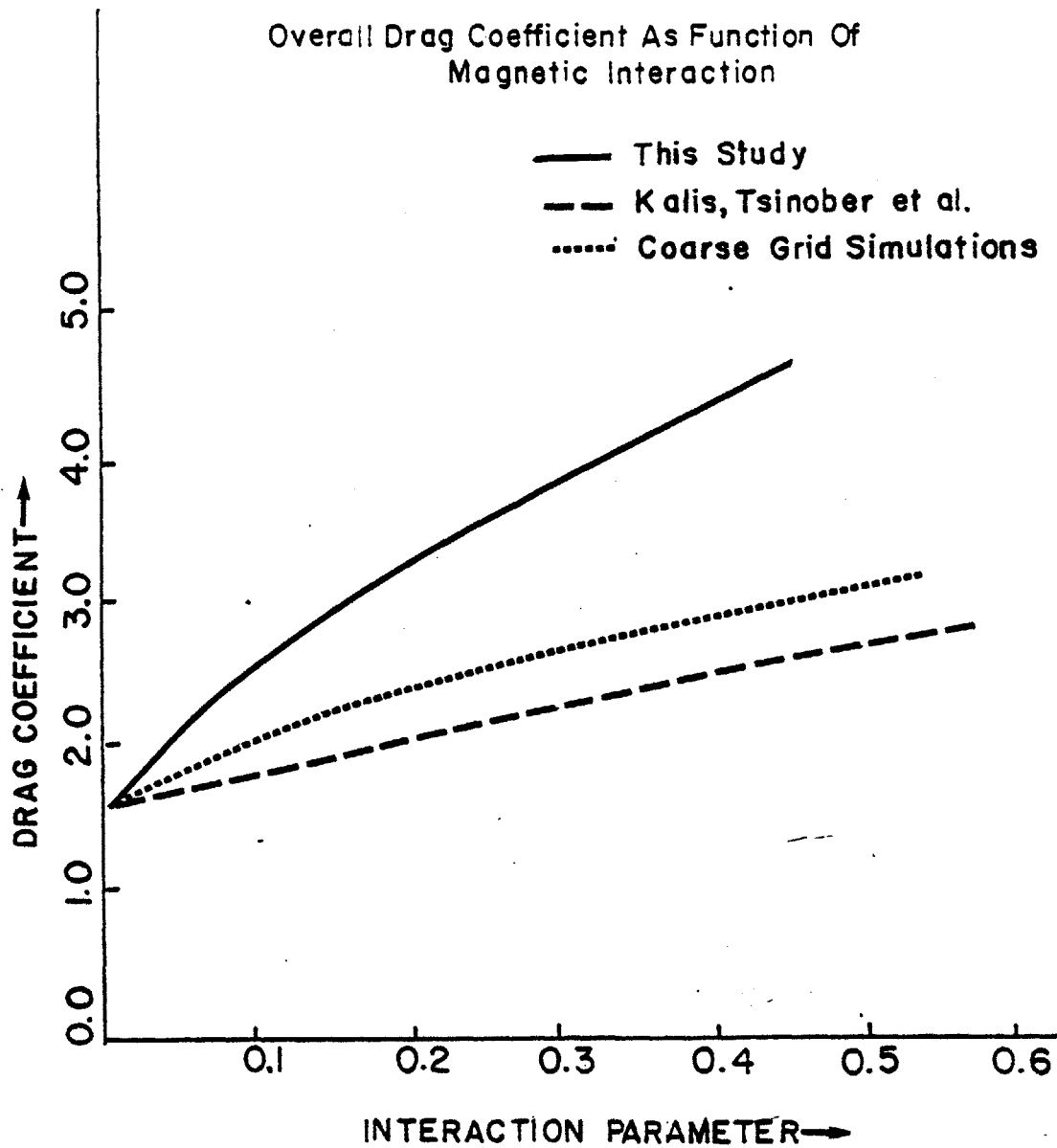


Fig. 5.41 Computed Variation of Drag Coefficient with Magnetic Interaction Parameter

Fig. 5.42 Example of Calibration Curve for Hot Film Anemometer - before Correction Analysis

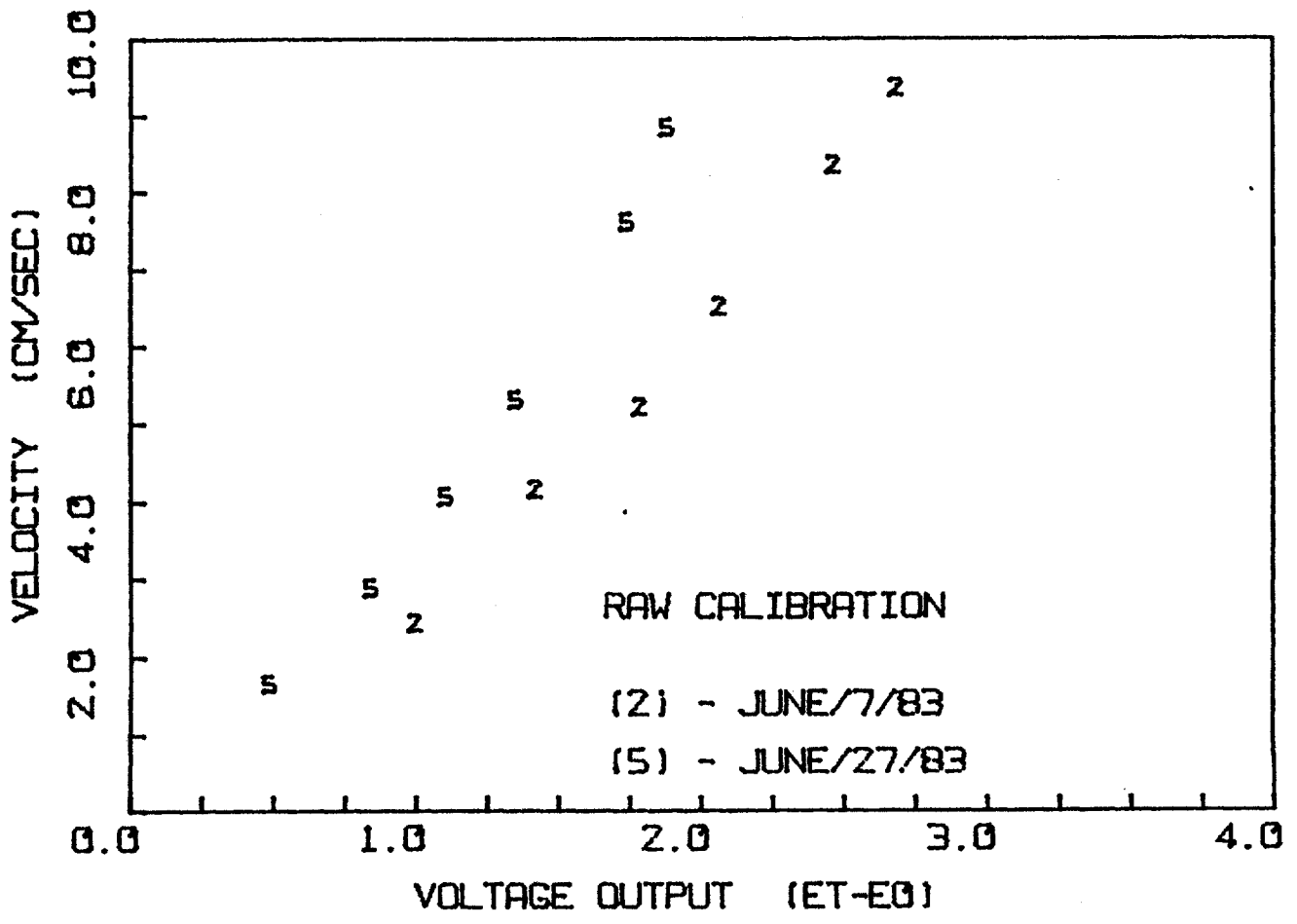
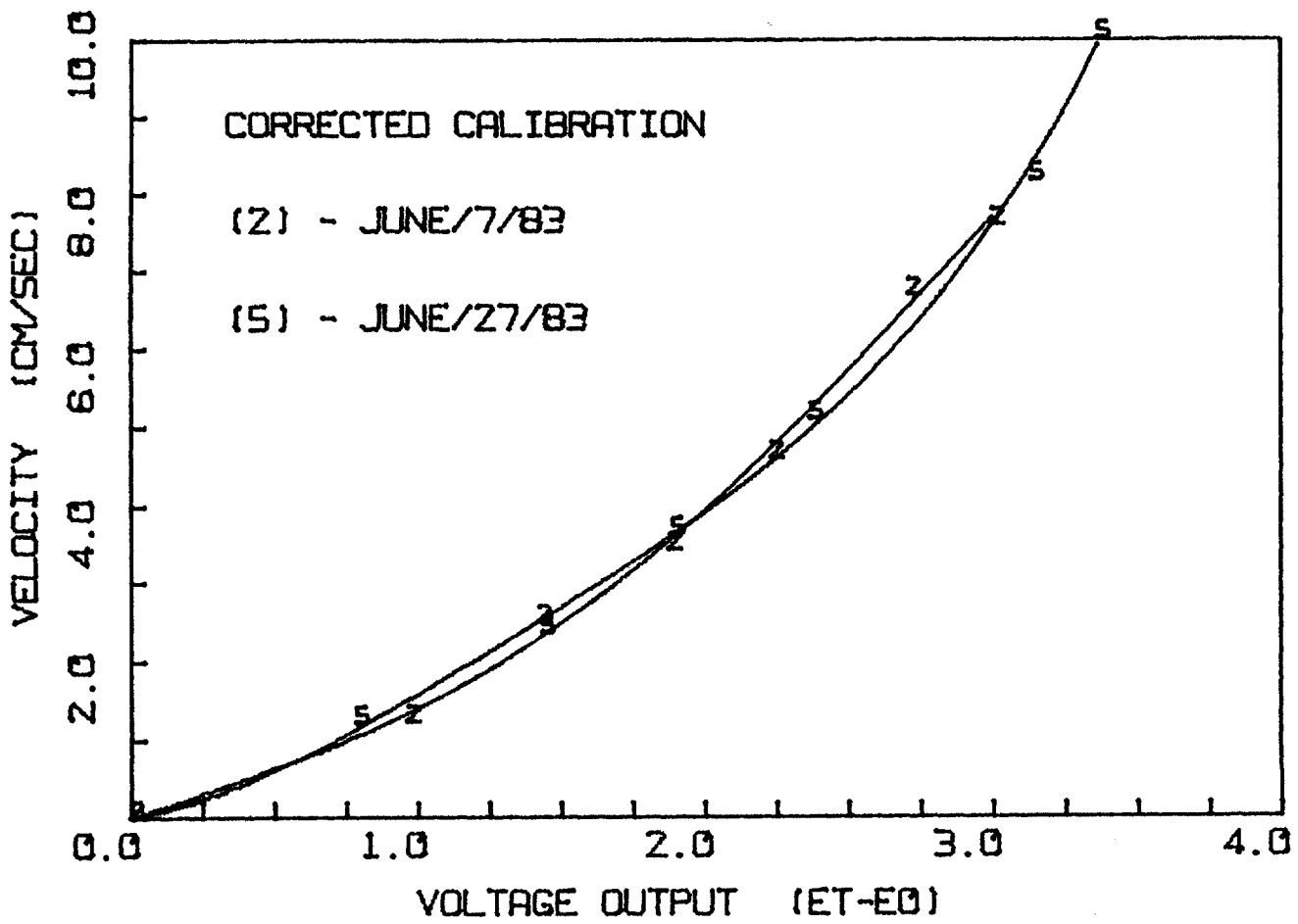


Fig. 5.43 Example of Same Calibration Curve - after correction analysis



## 6. CONCLUSIONS AND RECOMMENDATIONS

In this chapter some concluding remarks on the principal contributions and findings of this work are presented. Suggestions for further work along the lines of this investigation are also made.

### 6.1 Conclusions

Enumerated here are some of the contributions made by this work to the study of recirculating liquid metal flows in materials processing operations.

1) An experimental technique has been developed to make detailed measurements of flow and turbulence characteristics in a heated melt of liquid metal. The technique features modified probes for use in high temperature (upto 150° C) liquid metal melts, automatic calibration drift correction and a completely microcomputer controlled data acquisition and processing environment.

2) A mathematical model has been developed to represent the flow in the melt starting with calculation of the electromagnetic force field driving the flow and using the conventional  $\kappa - \epsilon$  model to represent the turbulence.



3) An alternative turbulence model has been proposed. This model, called the XI model, has been entirely developed as part of this work and has the following features:

(a) It is a computationally simple model that uses 1 transport equation to represent the turbulence as opposed to 2 transport equations for the  $\kappa - \epsilon$  model.

(b) It eliminates the need for most of the semi-empirical constants used by the  $\kappa - \epsilon$  model.

(c) It is intuitively appealing as it uses simple concepts like specific power input to the system and characteristic vessel size as important parameters to characterise the state of turbulence.

4) Comparison of predictions using the conventional models as well as the new XI model show that:

(a) The XI model calculation of mean flow as well as turbulence parameters is very close to those predicted by the  $\kappa - \epsilon$  model.

(b) Use of the Low Reynolds Number model does not result in any substantially different predictions.

(c) All three models predict the mean velocity and dissipation energy measured in the melt reasonably well. Turbulence intensity is, however, less well represented by all three models.

5) A parameter, called the 'Performance Index', (PI for short), is proposed. It is shown that this parameter can be used as a criterion for predicting the onset of turbulence in the recirculating flow in a melt more reliably than afforded by the use of a simple Reynolds Number evaluation.

6) Consideration of a 'Turbulence Efficiency' factor ( $\eta$ ) is proposed as an important criterion in determining the performance of a metallurgical agitation system. A calculation, made to evaluate this efficiency factor for gas stirred melts, shows  $\eta = 0.5$  in such systems. It is expected that this value of  $\eta$  will be different in different classes of systems.

7) A theory is proposed for explaining the variation of mixing times in metallurgical system reactors with different levels of power input and vessel size. The predicted variation using this theory is shown to match very well with the experimentally observed relationship. The fundamental basis of this theory, it is felt, permits its use in a wide variety of metallurgical systems to evaluate mixing or reaction times.

## 6.2 Suggestions for further work

This section presents some thoughts on how certain aspects of the work presented in this thesis may be extended to further enhance our knowledge of mixing in metallurgical systems.

1) It is felt that pulsating the applied current passing through the melt with a variable frequency presents possibilities for an interesting investigation into the turbulence, and thence, mixer performance of metallurgical reactors. If the frequency of pulsation is adjusted to be inside the spectral range of turbulence frequencies, alteration of the spectral shape and consequently, of the turbulence parameters, may result. Also, an oscillating force field may be expected to result in a (possibly) enhanced turbulence efficiency factor,  $\eta$ . This would have a direct effect on the turbulence levels and mixing times for the system. This line of investigation is presently being pursued as continuing graduate research work.

2) The XI model may be used for other turbulent recirculating flow problems and the results compared with similar computations using the  $\kappa - \epsilon$  model and experimental results, if available.

3) Values for turbulence efficiency need to be evaluated for different classes of metallurgical systems like the applied current configuration (eg. Electro Slag Refining) or the inductively stirred system (eg. Induction Furnace). This could be conveniently done using the same technique that was used for gas injected systems in this work, ie. by utilising information about the overall mixing time for the system.

APPENDIX A

## Thermal Anemometry Theory

The detecting element of a thermal anemometer consists of a very fine short metal wire (or film) which is heated by an electric current. The sensor is cooled by the flowing liquid, causing the heat transfer rate from the wire/film to increase. Typical sensors, of the hot wire and hot film type are shown below:

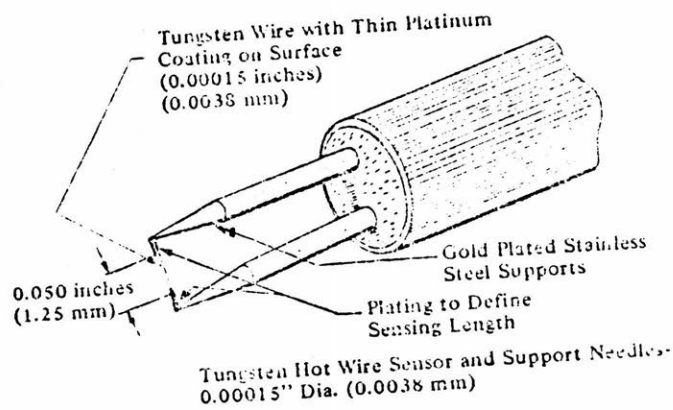


Fig. A.1

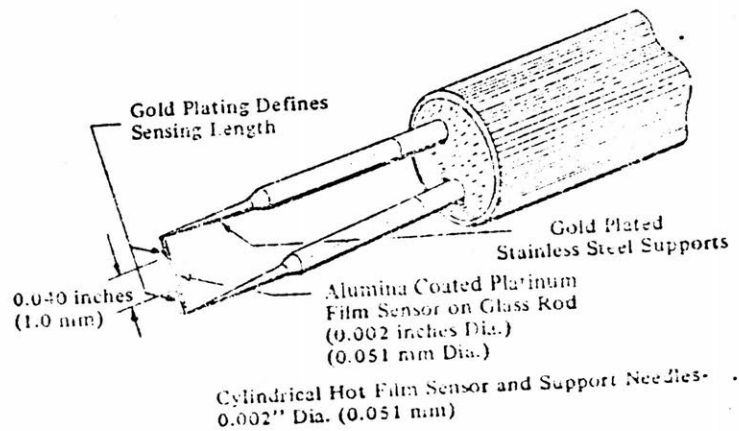
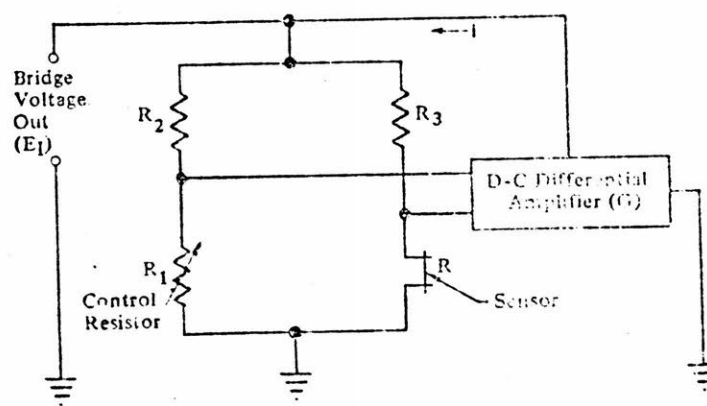


Fig. A.2

A very important part of the hot wire/film anemometer is the control circuit. There are two basic types of measurement schemes possible:

- (1) The Constant Current Anemometer
- and
- (2) The Constant Temperature Anemometer.

The constant current type of anemometer operates by taking the voltage signal caused by wire resistance changes and compensates for frequency lag with a non-linear amplifier. The constant temperature control scheme, while certainly not a recent innovation, has gained rapidly in acceptance during the last few years. It operates by utilising feedback controlled bridge circuit to maintain the sensor at constant temperature. The figure below depicts a constant temperature system.



Schematic of Constant Temperature System

Fig. A.3

As the velocity past the sensor increases the sensor will tend to cool with a resulting decrease in the resistance. This

resistance decrease will cause the voltage to decrease changing the input to the amplifier. The phase of the amplifier is such that this decrease in voltage will cause an increase in the output of the amplifier to increase the current through the sensor. If the amplifier has a sufficient gain, it will tend to keep its inputs very close to the balanced condition. Therefore any change in the sensor resistance will be immediately corrected by an increase or decrease in the current through the sensor. The output of the constant temperature system is the voltage output of the amplifier which in turn is the voltage required to drive the necessary current through the sensor. Since with feedback control the resistances in the bridge are constant, the voltage across the bridge is directly proportional to the current through the sensor and power is equal to current squared times the probe resistance. Therefore, the square of the voltage measured on top of the bridge is directly proportional to the instantaneous heat transfer between the sensor and environment.

APPENDIX B**Calculation of  $\kappa$  and  $\epsilon$  from the Spectrum**

A typical frequency domain spectrum is shown in Fig. 5.35 and represents a plot of  $E(n)$  versus  $n$ , where  $n$  is the frequency in Hertz. From this spectrum, the values of the Kinetic Energy of Turbulence,  $\kappa$ , and the Dissipation Rate of Turbulence,  $\epsilon$ , are calculated using the following relationships:

$$\kappa = \frac{3}{2} (GD \times GF)^2 \times \int_0^{\infty} E(n) dn$$

and

$$\epsilon = \frac{60\pi^2 \nu}{U^2} (GD \times GF)^2 \times \int_0^{\infty} n^2 E(n) dn$$

where  $GD$  is the gradient of the calibration curve at the operating point specified by the mean velocity in M/sec/Volt and  $GF$  is the gain factor set on the analog data recorder input amplifiers. The other symbols have their usual meanings.



APPENDIX C**Properties of Wood's Alloy****COMPOSITION:**

Bismuth	4 parts
Lead	2 part
Cadmium	1 part
Tin	1 part

**PHYSICAL<sup>(87)</sup> :**

Freezing Range	69 - 72 ° C
Density	$8.4 \times 10^3$ Kg/m <sup>3</sup>
Electrical Conductivity	$.99 \times 10^6$ mho/m
Molecular Viscosity	$2.29 \times 10^{-3}$ N/s/m <sup>2</sup>

APPENDIX D**Equipment Specifications:****MICROCOMPUTER AND PERIPHERICALS:**

- \* APPLE II Plus, 48 K RAM, MOSTEK 6502 Processor, 2 disk drives, BASIC and ASSEMBLER support
- \* AI13 12 Bit A/D Converter, 16 Channel, 20  $\mu$  s conversion time, (Interactive Structures, Bala Cynwyd, Pennsylvania)
- \* PL12 Flat Bed Plotter, 7" x 10" (also from Interactive Structures)
- \* Micromodem II, 300 Baud, Direct Connect Modem (Hayes Microcomputer Products, Norcross, Georgia)
- \* MPI - 88G Dot Matrix Printer (Microperipherals Inc., Salt Lake City, Utah)
- \* 'The Clock', Real time clock with 1msec BCD bits (Mountain Computer Inc., Santa Cruz, California)

**DC CURRENT GENERATOR:**

Gold Star 600SS, Rated 600 Amps at 60% Duty Cycle with remote contactor control. (Miller Electric Mfg. Co., Appleton, Wisconsin)

**SPECTRUM ANALYSIS:**

- \* HP 5423 A Structural Dynamics Analyser
- \* HP 54470 B Digital Filter
- \* HP 54410 B A/D Converter
- \* HP 9872 S Flat Bed Plotter

(All from Hewlett Packard Co., Palo Alto, California)

**THERMAL ANEMOMETRY:**

- \* (2) Series 1050 Constant Temperature
- \* (2) Series 1057 Signal Conditioner
- \* Series 1051 Monitor and Power Supply
- \* 1212-60 Hg. Probes

(All from Thermo Systems Inc., St. Paul, Minnesota)

**ANALOG DATA RECORDING:**

Teac R 61 FM Data Recorder, 4 Channel, Single Speed,  
Variable Gain (TEAC Corporation, Tokyo, Japan)

APPENDIX E

**Microcomputer Program Listings**

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100 TEXT : HOME : FLASH : HTAB 10: PRINT "RAW DATA ACQ": NORMAL
200 DIM A$(9),B(9),C(9),A(9)
220 GOSUB 9700: REM HIMEM SETTING
290 K = 0
300 ARYSZ = 36609 + K:STALOC = 36611 + K:ACTSTL = 36613 + K
310 ENLOC = 36615 + K:MULTSZ = 36617 + K:MLPNM = 36619 + K
320 SLPNM = 36621 + K:INERVL = 36623 + K:NMCHN = 36624 + K
325 GSTAL = 36625 + K:GNCD = 36646 + K:TYPE = 36651 + K
330 MEXCLN = 36655 + K:TIMEIN = 36656 + K:L2AP = 36662 + K
340 R1EF = 36668 + K:R2EF = 36669 + K:SEXCLN = 36671 + K
350 L1AP = 36672 + K:S1TAL = 36678 + K:CLSYS = 36730 + K
360 G1R = 36698 + K:XPND = 36702 + K:XCRD = 36704 + K
370 YCRD = 36706 + K:ZCRD = 36708 + K:RTN = 36710 + K
380 CURR = 36712 + K:TEMP = 36714 + K:OHR = 36716 + K:SVLTG = 36718
390 XDTE = 36732 + K:PARA = 36608 + K:HOT1 = 36736 + K
500 D$ = CHR$(4):R$ = CHR$(13):H = 256
1000 PRINT R$;D$;"BLOAD DAPARA,D1"
1010 INPUT " FIRST TIME TO THIS ROUT. (Y/N) ? ";AN$
1020 IF AN$ = "Y" THEN POKE 34,Z1: HOME : GOSUB 2000: REM VAR INPUT
1030 POKE 34,Z1: HOME : GOSUB 3000: REM VAR DISPLAY
1040 PRINT : INPUT " NEED TO CHANGE VARIABLES? Y/N ";AN$
1050 IF AN$ = "Y" THEN GOSUB 4000: GOTO 1030
1060 PRINT D$;"BLOAD HOT1.OBJO,D1"
1070 POKE 34, PEEK (37) - 1: HOME : INPUT " WHICH ROUTINE (MULT/SNGL) ";OP$
1080 POKE CHOICE, ASC ( LEFT$ (OP$,1)) + 128: POKE TYPE, ASC ( LEFT$ (OP$,1)):128
1085 IF LEFT$ (OP$,1) > < "S" THEN GOTO 1100
1090 POKE 36769,23: POKE 36888,0: POKE 36961,34. POKE 36989,234
1100 POKE 34, PEEK (37) - 1: HOME
1110 FLASH : PRINT "ARE YOU SURE THAT M/SW IS OFF(O V) Y/N?": GET AN$: NORMAL
1120 CALL HOT1: REM MAIN DATA ACQ ROUT
1125 POKE 36769,128: POKE 36888,17: POKE 36961,44: POKE 36989,96
1130 PRINT R$;D$;"RUN SAVDAT1,D1"
1140 END
2000 INVERSE : PRINT "VARIABLE INPUT MODE": NORMAL : POKE 34, PEEK (37)
2010 INPUT "NO. OF CHANNELS FOR SAMPLING=" ;C%: POKE NMCHN,C%
2020 FOR I = 1 TO C%
2030 PRINT I;"-CHN <GNCD, NO.> W/COMMA=" : VTAB PEEK (37): HTAB 33: INPUT D%,E%
2040 E% = E% + D% * 16: POKE GNCD + 5 - I,E%
2050 NEXT I
2060 INPUT "MAIN SWITCH CHANNEL NUMBER=" ;E%
2070 FOR J = I TO 5: POKE GNCD + 5 - J,E%: NEXT J
2080 INVERSE : PRINT "INPUT 4-HEX DIGITS PRECEDED BY '$' SIGN": NORMAL
2090 INPUT "DATA STACK SIZE=" ;A$(0)
2100 INPUT "ACT. START LOC. OF D-STACK=" ;A$(2)
2110 INPUT "START LOCATION OF D-STACK=" ;A$(1)
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2120 INPUT "MAX. END LOC. OF D-STACK= ";A$(3)
2130 INPUT "DATA SIZE OF MULT-ROUT= ";A$(4)
2140 INPUT "LOOP NO. OF MULT-ROUT= ";A$(5)
2150 INPUT "LOOP NO. OF SNGL-ROUT= ";A$(6)
2160 INPUT "LOOP NO. FOR INTERVAL= ";A$(7)
2170 FOR J = 0 TO 7
2180 A$ = A$(J): GOSUB 9800
2190 POKE ARYSZ - 1 + 2 * J,D: POKE ARYSZ + 2 * J,C
2200 NEXT J
2210 RETURN : END
3000 PRINT "A:NO.OF CHANNELS FOR SMPG="; TAB( 32); PEEK (NMCHN)
3010 FOR I = 1 TO PEEK (NMCHN)
3020 D% = INT ( PEEK (GNCD + 5 - I) / 16):E% = PEEK (GNCD + 5 - I) - D% * 16
3030 PRINT "A:";I;"-CHN GNCD="; TAB( 16);D%; TAB( 21);I;"-CHN NO.="; TAB( 36);E%
3040 NEXT I
3050 E% = PEEK (GNCD + 5 - I)
3060 PRINT "A:M/SW GNCD="; TAB( 16);"0"; TAB( 21);"M/SW CHN NO="; TAB( 36);E%: PRINT
3070 FOR J = 0 TO 7
3080 D = PEEK (ARYSZ - 1 + 2 * J):C = PEEK (ARYSZ + 2 * J): GOSUB 9900
3090 A$(J) = A$
3100 NEXT J
3110 PRINT "1:DATA ARY SZ="; TAB( 15);A$(0); TAB( 21);"2:START LOC="; TAB( 34);A$(1)
3120 PRINT "3:ACT.ST.LOC="; TAB( 15);A$(2); TAB( 21);"4:END LOC="; TAB( 34);A$(3)
3130 PRINT "5:MULT DAT.SZ="; TAB( 15);A$(4); TAB( 21);"6:L.NO.MULT="; TAB( 34);A$(5)
3140 PRINT "7:L.NO.SNGL="; TAB( 15);A$(6); TAB( 21);"8:L.NO.INTVL="; TAB( 34);A$(7): PRINT
3150 INVERSE : PRINT "SEE IF HIMEM =< ACTSTL, STALOC=ACTSTL-1"
3160 HTAB 7: PRINT "MULTSZ<$1000, ARYSZ<=$8FOO-ACTSTL"
3170 HTAB 7: PRINT "ARYSZ > ( 4 ) * NMCHN * LN(MULT/SNGL)": NORMAL
3180 RETURN : END
4000 Z2 = PEEK (37) - 1: POKE 34,Z2: HOME
4010 INPUT "TYPE ALPHA/NO./CTRL-S TO BE CHANGED ";C$
4020 IF LEFT$(C$,1) = CHR$(19) THEN GOTO 4180
4030 IF ASC (C$) = < 57 THEN GOTO 4150
4040 INPUT "NO. OF CHANNELS FOR SAMPLING= ";C%
4050 POKE NMCHN,C%
4060 FOR I = 1 TO C%
4070 PRINT I;"-CHN <GNCD, NO.> W/COMMA= ": VTAB PEEK (37): HTAB 33: INPUT D%.E%
4080 E% = E% + D% * 16: POKE GNCD + 5 - I,E%
4090 NEXT I
4100 INPUT "MAIN SWITCH CHANNEL NUMBER= ";E%
4110 FOR J = 1 TO 5
4120 POKE GNCD + 5 - J,E%
4130 NEXT J
4140 GOTO 4170
4150 INPUT "THEN VALUE(4-HEX DIG.W/'$')=? ";A$
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4160 GOSUB 9800:J = VAL (C$) - 1: POKE ARYSZ - 1 + 2 * J,D: POKE ARYSZ + 2 * J,C
4170 GOTO 4010
4180 RETURN : END
9700 C = PEEK (116):D = PEEK (115): GOSUB 9900
9710 VTAB 3: PRINT "PRESENT HIMEM="; TAB( 33);A$
9720 VTAB 4: HTAB 32: PRINT "=";D + C * 256:Z1 = PEEK (37)
9730 VTAB 5: PRINT "WANT TO CHANGE HIMEM Y/N?": VTAB 5: HTAB 32: GET AN$
9740 IF AN$ > < "Y" THEN GOTO 9780
9750 VTAB 5: HTAB 1: PRINT "INPUT NEW HIMEM "
9755 VTAB 5: HTAB 17: INVERSE : PRINT "(4-HEX;$XXXX)=": NORMAL : VTAB 5: HTAB 32: INPUT A$
9760 GOSUB 9800:B = C * 256 + D: VTAB 6: HTAB 32: PRINT "=";B
9770 HIMEM: B:Z1 = PEEK (37)
9780 POKE 34,Z1: HOME : RETURN : END
9800 FOR I = 2 TO 5
9810 A(I) = ASC ( MID$ (A$,I,1))
9820 IF A(I) = < 57 THEN A(I) = A(I) - 48: GOTO 9840
9830 A(I) = A(I) - 55
9840 NEXT I
9850 C = A(2) * 16 + A(3):D = A(4) * 16 + A(5)
9860 RETURN : END
9900 A(1) = INT (C / 16):A(2) = C - A(1) * 16:A(3) = INT (D / 16):A(4) = D - A(3) * 16:A$ = "$"
9910 FOR I = 1 TO 4
9920 IF A(I) < 10 THEN A$ = A$ + STR$ (A(I)): GOTO 9940
9930 A(I) = A(I) + 55:A$ = A$ + CHR$ (A(I))
9940 NEXT I
9950 RETURN : END
```

```
100 TEXT : HOME : FLASH : HTAB 7: PRINT "FULL CONV OF CAL-DATA": NORMAL
200 DIM A(9),B(9),C(19),D(9),E(9),G(9),GA(9)
210 GOSUB 9700: REM HIMEM SETTING
290 K = 0
300 ARYSZ = 36609 + K:STALOC = 36611 + K:ACTSTL = 36613 + K
310 ENLOC = 36615 + K:MULTSZ = 36617 + K:MLPNM = 36619 + K
320 SLPNM = 36621 + K:INERVL = 36623 + K:NMCHN = 36624 + K
325 GSTAL = 36625 + K:GNCD = 36646 + K:TYPE = 36651 + K
330 MEXCLN = 36655 + K:TIMEIN = 36656 + K:L2AP = 36662 + K
340 R1EF = 36668 + K:R2EF = 36669 + K:SEXCLN = 36671 + K
350 L1AP = 36672 + K:S1TAL = 36678 + K:CLSYS = 36730 + K
360 G1R = 36698 + K:XPNO = 36702 + K:XCRD = 36704 + K
370 YCRD = 36706 + K:ZCRD = 36708 + K:RTN = 36710 + K
380 CURR = 36712 + K:TEMP = 36714 + K:OHR = 36716 + K:SVLTG = 36718
390 XDTE = 36732 + K:PARA = 36608 + K:HOT1 = 36736 + K
500 D$ = CHR$(4):R$ = CHR$(13):H = 256
1000 POKE 34,Z1: HOME : PRINT : INPUT " DRIVE NO.? ":D%
1010 POKE 34,Z2: HOME : PRINT R$:D$:"CATALOG,D":D%
1020 INPUT " DATA ACQ TYPE =? (M/S) ":TY$
1030 INPUT " EXP. NO. = ":XP
1040 F$ = "C-" + STR$(XP) + "-" + TY$:G$ = F$ + "-TS"
1050 PRINT R$:D$:"BLOAD ":F$;" ,D":D%: HOME : VTAB 2: INVERSE : PRINT F$;" ,D":D%: NORMAL : POKE 34,3
1060 NC = PEEK (NMCHN):A% = 4:NP = 7
1070 GOSUB 6000: REM GAIN CODE CALC:(D(I)) ,(GA(I))
1080 IF TY$ > < "S" THEN GOTO 1100
1090 GOSUB 3000: GOTO 1110: REM SNGL CONV
1100 GOSUB 2000: REM MULT CONV
1110 GOSUB 4000
1120 INPUT " WANT TO CONV TS ? (Y/N) ":AN$
1130 IF LEFT$(AN$,1) > < "Y" THEN GOTO 1150
1140 PRINT R$:D$:"BLOAD ":G$;" ,D":D%: VTAB 2: INVERSE : PRINT G$;" ,D":D%: NORMAL : POKE 34,3: GOSUB 4500
1150 PRINT R$:D$:"RUN MENU,D1"
1160 END
2000 DS = PEEK (MULTSZ) * H + PEEK (MULTSZ - 1):DAS=DS:DBS=DS
2010 NBD = PEEK (SEXCLN) * H + PEEK (SEXCLN - 1):NAD = PEEK (MEXCLN) * H + PEEK (MEXCLN - 1)
2020 FOR I = 0 TO NC - 1
2030 A(I) = 0:B(I) = 0
2040 FOR K = 0 TO 4
2050 B(I) = B(I) + PEEK (S1TAL + 5 * I + K) * (H ^ K) / (DAS * NBD)
2060 A(I) = A(I) + PEEK (GSTAL + 5 * I + K) * (H ^ K) / (DAS * NAD)
2070 NEXT K
2080 B(I) = D(I) * B(I) / 4095 / GA(I):A(I) = D(I) * A(I) / 4095 / GA(I)
2090 NEXT I
2100 MSB = INT ( PEEK (L2AP + 5) / 16):MTB = PEEK (L2AP + 5) - MSB * 16
2110 TBM = PEEK (L2AP + 2) * 16 + PEEK (L2AP + 3) + PEEK (L2AP + 4) / 10 + MSB / 100 + MTB / 1000
```



```
2120 MSA = INT ( PEEK (L1AP + 5) / 16):MTA = PEEK (L1AP + 5) - MSA * 16
2130 TAM = PEEK (L1AP + 2) * 16 + PEEK (L1AP + 3) + PEEK (L1AP + 4) / 10 + MSA / 100 + MTA / 1000
2140 RETURN : END
3000 DBS = PEEK (MULTSZ) * H + PEEK (MULTSZ - 1):DAS = PEEK (SEXCLN) * H + PEEK (SEXCLN - 1)
3005 NBD = PEEK (MEXCLN) * H + PEEK (MEXCLN - 1):NAD = PEEK (SLPNM) * H + PEEK (SLPNM - 1)
3010 FOR I = 0 TO NC - 1
3020 A(I) = 0:B(I) = 0
3030 FOR K = 0 TO 4
3040 B(I) = B(I) + PEEK (GSTAL + 5 * I + K) * (H ^ K) / (DBS * NBD)
3050 A(I) = A(I) + PEEK (S1TAL + 5 * I + K) * (H ^ K) / (DAS * NAD)
3060 NEXT K
3070 B(I) = D(I) * B(I) / 4095 / GA(I):A(I) = D(I) * A(I) / 4095 / GA(I)
3080 NEXT I
3090 MSB = INT ( PEEK (L1AP + 5) / 16):MTB = PEEK (L1AP + 5) - MSB * 16
3100 TBM = PEEK (L1AP + 2) * 16 + PEEK (L1AP + 3) + PEEK (L1AP + 4) / 10 + MSB / 100 + MTB / 1000
3110 MSA = INT ( PEEK (L2AP + 5) / 16):MTA = PEEK (L2AP + 5) - MSA * 16
3120 TAM = PEEK (L2AP + 2) * 16 + PEEK (L2AP + 3) + PEEK (L2AP + 4) / 10 + MSA / 100 + MTA / 1000
3130 RETURN : END
4000 POKE 34,3: HOME
4010 INPUT " WANT TO PRINT OUT? (Y/N) ";AN$: IF LEFT$ (AN$,1) > < "Y" THEN HOME : GOTO 4030
4020 PRINT R$;D$;"PR#1": POKE 1401,80: HOME
4030 C(O) = PEEK (XDTE + 3) + PEEK (XDTE + 2) * H + PEEK (XDTE + 1) * H * H + PEEK (XDTE) * H * H * H
4040 A$ = STR$ (C(O))
4050 FOR J = 0 TO NP - 1:C(J) = PEEK (XCRD + 2 * J) * H + PEEK (XCRD + 2 * J + 1): NEXT J
4060 C(O) = C(O) / 1E3:C(5) = C(5) / 10:C(6) = C(6) / 100
4065 TV = C(O) / TAM
4070 FOR I = 1 TO 39: PRINT "*":: NEXT I: PRINT : PRINT
4080 PRINT "FILE NAME & EXP NO = ";TAB(20);F$;TAB(41);"EXPERIMENT DATE = ";LEFT$ (A$,4);"/";MID$ (A$,5,2);"/";RIGHT$ (A$,2)
4090 PRINT "CURR= ";C(4);" AMP"; TAB( 21);"OP TEMP= ";C(5);" DEG"; TAB( 41);"O/H R= ";C(6)
4110 PRINT "TR= ";C(O);" CM"; TAB( 21);"TV= ";TV;" CM/SEC"
4120 FOR J = 0 TO NC - 1
4130 PRINT "GNCD:CH-";J + 1;"=" ;G(J); TAB( 21);"REC GN:CH-";J + 1;"=" ;GA(J); TAB( 41);"SVLTG:CH-";J + 1;"=" ;E(J);" VOLT"
4140 PRINT "BASE VLTG: CH-";J + 1;"=" ;B(J);" VOLT"; TAB( 41);
4150 PRINT "ACTV VLTG: CH-";J + 1;"=" ;A(J);" VOLT"
4160 NEXT J
4170 PRINT "TOT SAMP TIM: BAS= ";TBM;" SEC"; TAB( 41);
4180 PRINT "TOT DAT PTS: BAS= ";DBS;"*";NBD;"=" ;DBS * NBD
4190 PRINT "TOT SAMP TIM: ACT= ";TAM;" SEC"; TAB( 41);
4200 PRINT "TOT DAT PTS: ACT= ";DAS;"*";NAD;"=" ;DAS * NAD
4210 PRINT : FOR I = 1 TO 39: PRINT "*":: NEXT I: PRINT
4220 PRINT R$;D$;"PR#0"
4230 RETURN : END
4500 POKE 34, PEEK (37): HOME : INPUT " WANT TO PRINT-OUT TS? (Y/N) ";AN$: IF LEFT$ (AN$,1) > < "Y" THEN GOTO 4520
4510 PRINT R$;D$;"PR#1": POKE 1401,80
4520 HOME : SP = PEEK (ACTSTL) * H + PEEK (ACTSTL - 1)
```

```
4530 FOR J = 0 TO NAD - 1
4540 PT = SP + J * A% * NC
4550 FOR I = 0 TO NC - 1
4560 B(I) = 0:PR = PT + I * A%
4570 FOR K = 0 TO A% - 1
4580 B(I) = B(I) + PEEK (PR + K) * (H ^ K) / DAS: NEXT K
4590 B(I) = B(I) * D(I) / 4095 / GA(I)
4600 NEXT I
4610 PRINT R$;J + 1;": ";
4620 FOR I = 0 TO NC - 1
4630 TP = I * 18 + 7:TP = TP - INT (TP / 40) * 40: IF PEEK (36) = > TP THEN TP = TP + 40
4640 PRINT TAB( TP);"C";I + 1;"="";B(I);
4650 NEXT I: NEXT J
4660 PRINT R$;D$;"PR#0": RETURN : END
6000 FOR I = 0 TO NC - 1
6010 G(I) = INT ( PEEK (GNCD + 4 - I) / 16):C(I) = PEEK (GNCD + 4 - I) - G(I) * 16:D(I) = G(I)
6020 IF D(I) > = 4 THEN D(I) = D(I) - 4
6030 B% = INT (D(I) / 2 + 0.5):C% = D(I) - INT (D(I) / 2) * 2
6040 D(I) = 5 * ((2 ^ C%) / (10 ^ B%))
6045 GA(I) = PEEK (G1R + I) / 100
6048 E(I) = PEEK (SVLTG + 2 * I) * H + PEEK (SVLTG + 2 * I + 1):E(I) = E(I) / 1E3
6050 NEXT I
6060 RETURN : END
9700 C = PEEK (116):D = PEEK (115):GOSUB 9900
9710 VTAB 3: PRINT "PRESENT HIMEM="; TAB( 33);A$
9720 VTAB 4: HTAB 32: PRINT "=";D + C * 256:Z1 = PEEK (37)
9730 VTAB 5: PRINT "WANT TO CHANGE HIMEM Y/N?": VTAB 5: HTAB 32: GET AN$
9740 IF AN$ > < "Y" THEN GOTO 9780
9750 VTAB 5: HTAB 1: PRINT "INPUT NEW HIMEM "
9755 VTAB 5: HTAB 17: INVERSE : PRINT "(4-HEX;$XXXX)=": NORMAL : VTAB 5: HTAB 32: INPUT A$
9760 GOSUB 9800:B = C * 256 + D: VTAB 6: HTAB 32: PRINT "=";B
9770 HIMEM: B:Z1 = PEEK (37)
9780 POKE 34,Z1: HOME : RETURN : END
9800 FOR I = 2 TO 5
9810 A(I) = ASC ( MID$ (A$,I,1))
9820 IF A(I) = < 57 THEN A(I) = A(I) - 48: GOTO 9840
9830 A(I) = A(I) - 55
9840 NEXT I
9850 C = A(2) * 16 + A(3):D = A(4) * 16 + A(5)
9860 RETURN : END
9900 A(1) = INT (C / 16):A(2) = C - A(1) * 16:A(3) = INT (D / 16):A(4) = D - A(3) * 16:A$ = "$"
9910 FOR I = 1 TO 4
9920 IF A(I) < 10 THEN A$ = A$ + STR$ (A(I)): GOTO 9940
9930 A(I) = A(I) + 55:A$ = A$ + CHR$ (A(I))
9940 NEXT I
```

FILE: CALFUL1 BAS

A1

VM/SP CONVERSATIONAL MONITOR SYSTEM

PAGE 004

9950 RETURN : END

```
100 TEXT : HOME : FLASH : HTAB 7: PRINT "POLY-FITTING OF CAL-CURVE": NORMAL
110 HIMEM: 36864: REM 36864=$9000
120 POKE 34,3: HOME
200 DIM X(100),Y(100),W(100),CY(100),E(100),C(10),A(10,11),XN(100)
300 D$ = CHR$(4):R$ = CHR$(13)
500 REM PROGRAM FOR POLYNOMIAL FITTING TO A SET OF DATA
510 REM BY LEAST-SQUARE-METHOD
520 GOSUB 4000: REM FILE READING SUBR
530 GOSUB 4500: REM ORIG. COEFF. MATRIX MAKING SUBR
540 PRINT : PRINT "***ORIGINAL SQUARE COEFFICIENT MATRIX***": PRINT : GOSUB 2500: REM MATRIX PRINTING SUBR
550 GOSUB 1000: REM LOWER/UPPER TRIANGULAR MATRIX DECOMPOSITION
560 PRINT : PRINT "***L/U DECOMPOSED MATRIX***": PRINT : GOSUB 2500
570 REM RESET THE RHS INTO C (COEFF)--DO THIS FOR EACH DEGREE
580 FOR I = (MS + 1) TO SF
590 PRINT : PRINT "*****POLYNOMIAL---DEGREE ";I - 1;"*****"
600 FOR J = 1 TO I:C(J) = A(J,TF): NEXT J
610 GOSUB 2000: REM SOLUTION SUBROUTINE
620 PRINT : PRINT "***FINAL RESULT:COEFFICIENTS***": PRINT
630 GOSUB 2700: REM COEFF PRINTING SUBR
640 GOSUB 5000: REM REFERENCE OUTPUT SUBR
650 GOSUB 5500: REM COEFF-FILE SAVE
660 NEXT I: REM TRY NEXT DEGREE OF POLYNOMIAL
670 PRINT R$;D$:"RUN MENU.D1"
680 END
1000 REM THIS SUBR FORMS L/U EQUIVALENT OF SQ COEFF MATRIX A
1010 REM THE L/U MATRIX,IN COMPACT FORM, IS RETURNED IN A
1020 FOR I = 1 TO SF
1030 FOR J = 2 TO SF
1040 SUM = 0
1050 IF J > I THEN GOTO 1110
1060 FOR K = 1 TO J - 1:SUM = SUM + A(I,K) * A(K,J): NEXT K
1070 A(I,J) = A(I,J) - SUM
1080 GOTO 1170
1110 IF (I - 1) = 0 THEN GOTO 1150
1120 FOR K = 1 TO I - 1:SUM = SUM + A(I,K) * A(K,J): NEXT K
1130 REM TEST FOR SMALL VALUE ON DIAGONAL
1150 IF ABS(A(I,I) < 1.OE - 10) THEN GOTO 1200
1160 A(I,J) = (A(I,J) - SUM) / A(I,I)
1170 NEXT J
1175 NEXT I
1180 RETURN : END
1200 PRINT "REDUCTION NOT COMPLETED, DUE TO SMALL VALUE IN DIVISOR IN ROW-";I: RETURN
1210 END
2000 REM THIS SUBR FINDS SOLUTION TO A SET OF I-LINERAR EQUATION
2010 REM THAT CORRESPONDS TO THE RHS VECTOR C.
```

```
2020 REM A-MATRIX =L/U DECOMPSION EQUIV TO ORIG COEFF MATRIX
2030 REM BY L/U REDUCTION PROCEDURE, SOLUTION VECTOR IS RETURNED IN C VECTOR
2040 REM NOW, DO THE REDUCTION STEP
2050 C(1) = C(1) / A(1,1)
2060 FOR J = 2 TO I
2070 SUM = 0
2080 FOR K = 1 TO J - 1:SUM = SUM + A(J,K) * C(K): NEXT K
2090 C(J) = (C(J) - SUM) / A(J,J)
2100 NEXT J
2102 PRINT : PRINT "***FINALLY DECOMPOSED LAST COLUMN OF L/U MATRIX***": PRINT
2104 GOSUB 2700
2110 REM NOW,DO BACK SUBSTITUTION,AND DIAGONAL ELEMENTS OF U-MATRIX ARE ALL 1S.
2120 FOR J = 2 TO I
2130 SUM = 0
2140 FOR K = (I - J + 2) TO I:SUM = SUM + A(I - J + 1,K) * C(K): NEXT K
2150 C(I - J + 1) = C(I - J + 1) - SUM
2155 NEXT J
2160 RETURN
2170 END
2500 REM (N X N+1) MATRIX PRINTING SUBR
2510 FOR I = 1 TO SF
2520 PRINT "<";I;">-TH ROW ELEMENTS"
2530 FOR J = 1 TO TF
2532 TB = 20 * (J - 1) + 1:TB = TB - INT (TB / 40) * 40
2534 IF PEEK (36) = > TB THEN TB = TB + 40
2540 PRINT TAB( TB);"(";J;" ) ";A(I,J);
2550 NEXT J
2560 PRINT : NEXT I
2570 RETURN
2580 END
2700 REM COEFF PRINTING SUBR
2710 FOR J = 1 TO I:TB = 20 * (J - 1) + 1:TB = TB - INT (TB / 40) * 40
2720 IF PEEK (36) = > TB THEN TB = TB + 40
2730 PRINT TAB( TB);"(";J;" ) ";C(J);: NEXT J: PRINT
2740 RETURN
2750 END
4000 REM FILE READING SUBROUTINE
4010 INPUT "REF'NED CAL-ASSY F.N. = ";F$
4020 PRINT R$;D$;"OPEN ";F$,".D2"
4040 PRINT R$;D$;"READ ";F$
4045 INPUT EO,SZ
4050 FOR I = 1 TO SZ
4060 INPUT X(I),Y(I)
4065 W(I) = 1
4070 NEXT I
```

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```
4080 PRINT R$;D$;"CLOSE ";F$
4090 N = SZ: PRINT "*NO. OF POINTS: N = ";N
4095 IX = EO
4098 FOR I = 1 TO N:X(I) = X(I) - IX: NEXT I
4100 PRINT : INPUT "STARTING, FINAL DEG OF POLYNOM-APPROX:(DEG. LE. N-1 & .LE. 9) =? ";MS,MF
4110 SF = MF + 1:TF = MF + 2
4120 RETURN : END
4500 REM ORIGINAL COEFFICIENT MATRIX MAKING SUBR(N ROWS,N+1 COL)
4510 REM ARRAY XN HOLDS POWER OF THE EACH X VALUE
4520 FOR I = 1 TO N:XN(I) = 1: NEXT I
4530 REM COMPUTE 1ST / N+1ST COLUMN OF A.
4540 REM I MOVES DOWN THE ROWS, J SUMS OVER THE N VALUES
4550 FOR I = 1 TO SF
4560 A(I,1) = 0:A(I,TF) = 0
4570 FOR J = 1 TO N
4580 A(I,1) = A(I,1) + XN(J) * W(J)
4590 A(I,TF) = A(I,TF) + Y(J) * XN(J) * W(J)
4600 XN(J) = XN(J) * X(J)
4610 NEXT J
4620 NEXT I
4630 REM COMPUTE THE LAST ROW OF A. I (COLUMN #). J SUMS OVER N VALUES
4640 FOR I = 2 TO SF
4650 A(SF,I) = 0
4660 FOR J = 1 TO N
4670 A(SF,I) = A(SF,I) + XN(J) * W(J)
4680 XN(J) = XN(J) * X(J)
4690 NEXT J
4700 NEXT I
4710 REM FILL THE REST OF A. I-ROW,J MOVES ACROSS THE COLUMNS
4720 FOR J = 2 TO SF
4730 FOR I = 1 TO MF
4740 A(I,J) = A(I + 1,J - 1)
4750 NEXT I
4760 NEXT J
4770 RETURN : END
5000 REM REFERENCE OUTPUT SUBROUTINE
5010 VAR = 0:ESQ = 0
5020 FOR J = 1 TO N
5030 CY(J) = 0
5040 FOR K = 2 TO I:CY(J) = (CY(J) + C(I - K + 2)) * X(J): NEXT K
5050 CY(J) = CY(J) + C(1):ER = Y(J) - CY(J)
5055 IF Y(J) = 0 THEN E(J) = 999999999: GOTO 5060
5058 E(J) = ER * 100 / Y(J)
5060 ESQ = ESQ + ER ^ 2
5070 NEXT J
```

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```
5080 VAR = ESQ / (N - 1)
5090 PRINT : PRINT "SUM OF ERROR SQ'ED = ";ESQ; SPC( 5);"VARIANCE OF FITTING = ";VAR: PRINT
5095 PRINT "NO. "; TAB( 10);"X"; TAB( 19);"Y"; TAB( 30);"CY"; TAB( 45);"ERROR, %"
5098 FOR ZZ=1 TO 79:PRINT "-";:NEXT ZZ:PRINT
5100 FOR J = 1 TO N
5110 J$ = STR$(J)
5120 IF INT (J / 10) = 0 THEN J$ = " " + J$
5130 PRINT J$; TAB( 5);X(J); TAB( 18);Y(J); TAB( 26);CY(J); TAB( 49);E(J)
5140 NEXT J
5150 RETURN : END
5500 INPUT "CAL-COEFF FILE NAME = ";G$
5510 PRINT R$;D$;"OPEN ";G$;".D2"
5520 PRINT R$;D$;"DELETE ";G$
5530 PRINT R$;D$;"OPEN ";G$
5540 PRINT R$;D$;"WRITE ";G$
5550 PRINT C(1);".";C(2);".";C(3);".";C(4);".";C(5)
5560 PRINT EO
5570 PRINT R$;D$;"CLOSE ";G$
5580 RETURN : END
```

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SOURCE FILE: HOT1.....

```

1 *****
0000: 2 *THIS ASSEMBLY ROUTINE CONSISTS OF TWO MAIN PARTS (GETMULT
0000: 3 * & GETSNGL ), PREPARATION STEP AND OTHER SUBROUTINES.
4 *****

0000: 6 *ZERO PAGE VARIABLES

0006: 8 ZLOC EQU $06 * ZERO LOCATION PTR,DUMMY,=1B,ASM
0007: 9 ZPAGE EQU $07 * PAGE COUNTER(DATA ARRAY),=1B,ASM
0008: 10 XSAV1 EQU $08 * X REGISTER SAVE1,=1B,ASM
0009: 11 XSAV2 EQU $09 * X REGISTER SAVE2,=1B,ASM
0019: 12 ZNMCHN EQU $19 * NO. OF CHANNELS,=1B,ASM
001A: 13 CHN1V EQU $1A * 1ST CHAN VLTG(L),3B>,ASM
001D: 14 CHN2V EQU $1D * 2ND CHAN VLTG(L),3B>,ASM
00CF: 15 ZLPNUM EQU $CF * LOOPING NO.(H),<2B,ASM
00EB: 16 ZGNCD EQU $EB *GAINCODE(G5,G4,G3,G2,G1),5B>,ASM
00F9: 17 YSAV1 EQU $F9 * Y REGISTER SAVE1,=1B,ASM
00FA: 18 CHN3V EQU $FA * 3RD CHAN VLTG(L),3B>,ASM
00FD: 19 CHN4V EQU $FD * 4TH CHAN VLTG(L),3B>,ASM

0000: 21 *$8F PAGE VARIABLES

8F01: 23 ARYSZ EQU $8F01 **DATA ARRAY SIZE(H),<2B,BASIC
8F03: 24 STALOC EQU $8F03 **START LOC. OF D.ARY(H),ACT.LOC-1
8F05: 25 ACTSTL EQU $8F05 **ACT.START LOC OF D.ARY(H),<2B,BAS
8F07: 26 ENDLOC EQU $8F07 * END LOCATION OF DAT ARY(H),<2B,ASM
8F09: 27 MULTSZ EQU $8F09 **SIZE OF MULT DATA(H),<2B,BASIC
8F0B: 28 LNMULT EQU $8F0B **LOOPING NO. OF MULT(H),<2B,BASIC
8F0D: 29 LNSNGL EQU $8F0D **LOOPING NO. OF SNGL(H),<2B,BASIC
8F0F: 30 INTVL EQU $8F0F **LP NO. OF INTERVAL DELAY(H),<2B
8F10: 31 NMCHN EQU $8F10 **NO.OF CHANNELS,=1B,BASIC
8F11: 32 GTOT1 EQU $8F11 * GRAND TOTAL1(L),5B>,ASM
8F16: 33 GTOT2 EQU $8F16 * GRAND TOTAL2(L),5B>,ASM
8F1B: 34 GTOT3 EQU $8F1B * GRAND TOTAL3(L),5B>,ASM
8F20: 35 GTOT4 EQU $8F20 * GRAND TOTAL4(L),5B>,ASM
8F25: 36 CHOICE EQU $8F25 * CHOICE DISPLAY,=1B,ASM
8F26: 37 GNCD EQU $8F26 **GAINCODE,(G5,G4,G3,G2,G1),5B>,BASIC
8F3C: 38 REFTIM1 EQU $8F3C *REF. TIME1,=1B,ASM
8F3D: 39 REFTIM2 EQU $8F3D *REF. TIME2,=1B,ASM
8F2C: 40 DLN1 EQU $8F2C * DISP.L.N.(1,2 * NMCHN FOR S,M)
8F2D: 41 DLN2 EQU $8F2D * DISP.L.NO.(2,4 FOR S,M),=1B,ASM
8F2F: 42 EXCMLN EQU $8F2F * EXEC'TED LP.NO OF MULT
8F3F: 43 EXCSLN EQU $8F3F * EXEC'TED LP NO OF SNGL
8F36: 44 LAPTIM2 EQU $8F36 * LAP TIME,6B>,ASM
    
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8F30:      45 TIMEIN EQU $8F30      * TIME INITIAL,6B>,ASM
8F40:      46 LAPTIM1 EQU $8F40     * LAP TIME,6B>,ASM
8F46:      47 STOT1 EQU $8F46       *SUBTOTAL1(L),5B>,ASM
8F4B:      48 STOT2 EQU $8F4B       *SUBTOTAL2(L),5B>,ASM
8F50:      49 STOT3 EQU $8F50       *SUBTOTAL3(L),5B>,ASM
8F55:      50 STOT4 EQU $8F55       *SUBTOTAL4(L),5B>,ASM
8F80:      51 ORIGIN EQU $8F80      * ORIGIN OF THIS BINARY FILE

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0000:      53 *OTHER PAGE VARIABLES

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COCO:      55 CLOCK EQU $COCO       *CLOCK, 5B>, ASM
COD0:      56 AI13 EQU $COD0        *ANAL/DIG CONV, 3B>, ASM

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0000:      58 *****
          59 *PREPARATION SECTION
          60 *****

```

```

----- NEXT OBJECT FILE NAME IS HOT1.OBJO

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8F80:      62          ORG ORIGIN
8F80:A9 00  63          LDA #$00      *ZLOC=$00
8F82:85 06  64          STA ZLOC
8F84:20 16 92 65          JSR PARYCLR  *G/S-TOT & EXC(M/S)LN CLEAR
8F87:20 50 92 66          JSR TCARYCL  *INIT/LAP TIME CHNV CLR
8F8A:AD 10 8F 67          LDA NMCHN   *ZNMCHN=NMCHN
8F8D:85 19  68          STA ZNMCHN

8F8F:20 7C 92 70          JSR SCRNCLE
8F92:A9 01  71          LDA #$01      *VTAB=01
8F94:85 22  72          STA $22
8F96:AD 25 8F 73          LDA CHOICE  *OPTION(MULT OR SNGL)
8F99:C9 D3  74          CMP #'S'
8F9B:FO 03  75          BEQ SKIP77
8F9D:4C A3 8F 76          JMP BSVLTG
8FA0:4C 78 90 77 SKIP77  JMP GETSNGL

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8FA3:      79 *****
          80 *MULTIPLE DATA PICKING ROUTINE
          81 *****

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```

8FA3:      83 *BASE VOLTAGE MEASURE ROUTINE

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170
8FA3:A2 00      85 BSVLTG LDX  # $00      *CAPTION 1 DISP & CHOICE
8FA5:20 84 92   86          JSR  HDLNCHO

8FA8:A2 05      88          LDX  # $05      *ZGNCD=GNCD
8FAA:BD 25 8F   89 LOOP72 LDA  GNCD-1,X
8FAD:95 EA      90          STA  ZGNCD-1,X
8FAF:CA         91          DEX
8FBO:DO F8      92          BNE  LOOP72

8FB2:20 30 92   94          JSR  DARYCLR  *DATA ARRAY CLEAR
8FB5:20 F3 94   95          JSR  PTRSET

8FB8:A0 00      97          LDY  # $00      *LNMULT
8FBA:20 8E 92   98          JSR  LPLMTST  *LOOPING LIMIT SET

8FBD:20 65 92   100         JSR  MSWAIT   *MAIN SW(O/5V) WAITING & BELL

8FC0:20 D4 92   102         JSR  INITIM   *INITIAL TIME SET SUBR

8FC3:20 01 91   104 LOOP70 JSR  MULTDAT
8FC6:20 9D 92   105         JSR  STOPTM
8FC9:20 1F 93   106         JSR  STORE    *STORE DATA INTO DATA ARRAY

8FCC:AD 25 8F   108         LDA  CHOICE
8FCF:C9 D9      109         CMP  #'Y'
8FD1:DO 03      110         BNE  EXIT74
8FD3:20 0E 94   111         JSR  DISPLY

8FD6:20 6E 92   113 EXIT74 JSR  MSWCHK   *MAIN SW CHECK STEP
8FD9:FO 08      114         BEQ  EXIT75

8FDB:C6 CE      116         DEC  ZLPNUM-1 *LOOPING TIME COUNT
8FDD:DO E4      117         BNE  LOOP70
8FDF:C6 CF      118         DEC  ZLPNUM
8FE1:10 E0      119         BPL  LOOP70

8FE3:20 9D 92   121 EXIT75 JSR  STOPTM
8FE6:20 FE 94   122         JSR  BELL
8FE9:20 C8 92   123         JSR  LAPTMV

8FEC:20 7C 92   125         JSR  SCRNCLE
8FEF:A2 22      126         LDX  # $22      *CAPTION-4 DISPLAY
8FF1:20 13 93   127         JSR  CAPTDISP

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8FF4:A0 00    129      LDY  #$00    *LNMULT
8FF6:A2 10    130      LDX  #$10    *EXCSLN
8FF8:20 5E 94  131      JSR  EXLPNM  *EXEC'TED LP NO CALC

8FFB:A0 34    133      LDY  #$34    *EXCSLN
8FFD:A2 35    134      LDX  #$35    *STOT
8FFF:86 09    135      STX  XSAV2
9001:20 7A 94  136      JSR  MULTADD

9004:20 C6 93  138      JSR  ARYDISP

9007:A2 35    140      LDX  #$35    *STOT
9009:20 EE 93  141      JSR  GTOTLAPT *STOT &LAPTIME & BELL
900C:EA EA EA  142      DFB  $EA,$EA,$EA

900F:          144 *ACTIVE VOLTAGE MEASURE ROUTINE

900F:A2 11    146 GETMULT LDX  #$11    *CAPTION 3 DISP & CHOICE
9011:20 84 92  147      JSR  HDLNCHO

9014:A2 05    149      LDX  #$05    *ZGNCD=GNCD
9016:BD 25 8F  150 LOOP12 LDA  GNCD-1,X
9019:95 EA    151      STA  ZGNCD-1,X
901B:CA      152      DEX
901C:DO F8    153      BNE  LOOP12

901E:20 30 92  155      JSR  DARYCLR *DATA ARRAY CLEAR
9021:20 F3 94  156      JSR  PTRSET

9024:A0 00    158      LDY  #$00    *LNMULT
9026:20 8E 92  159      JSR  LPLMTST *LOOPING LIMIT SET SUBR

9029:20 65 92  161      JSR  MSWAIT  *MAIN SW(O/5V) WAITING & BELL

902C:20 D4 92  163      JSR  INITIM  *INITIAL TIME SET SUBR

902F:20 01 91  165 LOOP20 JSR  MULTDAT *MULTIPLE DATA SAMPLING SUBR

9032:20 9D 92  167      JSR  STOPTM  *LAP TIME SETTING SUBR

9035:20 1F 93  169      JSR  STORE   *STORE DATA INTO DATA ARRAY

9038:AD 25 8F  171      LDA  CHOICE
903B:C9 D9    172      CMP  #'Y'
903D:DO 03    173      BNE  EXIT14

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```

903F:20 OE 94 174            JSR  DISPLY    *DISPLAY DATA ON SCRN

9042:20 6E 92 177 EXIT14    JSR  MSWCHK    *MAIN SW CHECK STEP
9045:FO 08 178            BEQ  EXIT15

9047:C6 CE 180            DEC  ZLPNUM-1  *LOOPING TIME COUNT
9049:DO E4 181            BNE  LOOP20
904B:C6 CF 182            DEC  ZLPNUM
904D:10 EO 183            BPL  LOOP20

904F:20 9D 92 185 EXIT15    JSR  STOPTM    *LAP TIME SETTING SUBR
9052:20 FE 94 186            JSR  BELL

9055:20 7C 92 188            JSR  SCRNCLE  *CLEAR SCRN
9058:A2 2C 189            LDX  #$2C     *CAPTION-5 DISPLAY
905A:20 13 93 190            JSR  CAPTDISP

905D:A0 00 192            LDY  #$00     *LNMULT
905F:A2 00 193            LDX  #$00     *EXCMLN
9061:20 5E 94 194            JSR  EXLPNM   *EXEC'TED LP NO CALC

9064:A0 24 196            LDY  #$24     *EXCMLN
9066:A2 00 197            LDX  #$00     *GTOT
9068:86 09 198            STX  XSAV2
906A:20 7A 94 199            JSR  MULTADD  *MULT DATA ADD SUBR

906D:20 C6 93 201            JSR  ARYDISP

9070:A2 00 203            LDX  #$00     *GTOT
9072:20 EE 93 204            JSR  GTOTLAPT *GTOT & LAPTIME & BELL
9075:60 205            RTS
9076:EA EA 206            DFB  $EA,$EA

9078:                    208 *****
                         209 *SINGLE DATA PICKING ROUTINE
                         210 *****

9078:A2 36 212 GETSNGL LDX  #$36     *CAPTION-6 DISPLAY & CHOICE
907A:20 84 92 213            JSR  HDLNCHO
    
```

FILE:	HOT1	ASSM	D1	VM/SP CONVERSATIONAL MONITOR SYSTEM	PAGE 006
907D:20	30	92	215	JSR DARYCLR	*DATA ARRAY CLEAR SUBR
9080:A6	19		217	LDX ZNMCHN	*ZGNCD=GNCD,DIFFERENT SEQ FROM MULT
9082:A0	04		218	LDY #\$04	
9084:B9	26	8F	219	LOOP32 LDA GNCD,Y	
9087:95	EB		220	STA ZGNCD,X	
9089:88			221	DEY	
908A:CA			222	DEX	
908B:10	F7		223	BPL LOOP32	
908D:A0	00		225	LDY #\$00	*LNMULT
908F:20	8E	92	226	JSR LPLMTST	*LOOPING LIMIT SET SUBR
9092:20	65	92	228	JSR MSWAIT	*MAIN SW(O/5V) WAITING & BELL
9095:20	D4	92	229	JSR INITIM	*INITIAL TIME SETTING SUBR
9098:29	FO		230	AND #\$FO	*4LSB-MASKING OF CLOCK+3
909A:8D	3D	8F	231	STA REFTIM2	*1-15 SEC
909D:20	F3	94	233	LOOP41 JSR PTRSET	
90A0:AE	0D	8F	235	LDX LNSNGL	*CHN4V+2=LNSNGL(H)
90A3:86	FF		236	STX CHN4V+2	
90A5:AE	0C	8F	237	LDX LNSNGL-1	*CHN4V+1=LNSNGL(L)
90A8:DO	O2		238	BNE SKIP31	
90AA:C6	FF		239	DEC CHN4V+2	
90AC:86	FE		240	SKIP31 STX CHN4V+1	
90AE:AE	0F	8F	242	LDX INTVL	*XSAV2=INTVL(H)
90B1:86	09		243	STX XSAV2	
90B3:AE	0E	8F	244	LDX INTVL-1	
90B6:86	08		245	STX XSAV1	
90B8:20	C8	91	247	JSR SNGLDAT	
90BB:AD	25	8F	249	LDA CHOICE	
90BE:C9	D9		250	CMP #'Y'	
90C0:DO	O3		251	BNE EXIT33	
90C2:20	B2	93	252	JSR LNDISP	*EXEC'TED LP NO DISP
90C5:20	FF	92	254	EXIT33 JSR TMSTEP2	*TIME STEP CHECK SUBR
90C8:20	6E	92	256	JSR MSWCHK	*MAIN SW VLTG CHK SUBR
90CB:FO	08		257	BEQ EXIT22	

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90CD:C6 CE      259      DEC  ZLPNUM-1
90CF:DO CC      260      BNE  LOOP41
90D1:C6 CF      261      DEC  ZLPNUM
90D3:10 C8      262      BPL  LOOP41

90D5:20 9D 92   264  EXIT22 JSR  STOPTM  *LAPTIME SETTING
90D8:20 FE 94   265      JSR  BELL
90DB:20 C8 92   266      JSR  LAPTMV

90DE:20 7C 92   268      JSR  SCRNCLE
90E1:A2 47      269      LDX  #$47  *CAPTION-7 DISPLAY LOOP
90E3:20 13 93   270      JSR  CAPTDISP

90E6:A0 00      272      LDY  #$00  *LNMULT
90E8:A2 10      273      LDX  #$10  *EXCSLN
90EA:20 5E 94   274      JSR  EXLPNM *EXEC'TED LOOPING NO CALC

90ED:A0 02      276      LDY  #$02  *LNSNGL
90EF:A2 35      277      LDX  #$35  *STOT
90F1:86 09      278      STX  XSAV2
90F3:20 7A 94   279      JSR  MULTADD *STOT ADD SUBR

90F6:20 C6 93   281      JSR  ARYDISP *ARRAY DISPLAY SUBR

90F9:A2 35      283      LDX  #$35  *STOT
90FB:20 EE 93   284      JSR  GTOTLAPT *STOT & LAP TIME DISP & BELL

90FE:60      286      RTS
90FF:EA EA      287      DFB  $EA,$EA

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9101:      289 *****
          290 *OTHER SUBROUTINES AND CONSTANTS
          291 *****

```

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9101:      293 *MULTIPLE DATA GETTING SUBROUTINE

9101:AE 09 8F   295 MULTDAT LDX  MULTSZ  *XSAV2= MULTSZ(H)
9104:86 09      296      STX  XSAV2
9106:AE 08 8F   297      LDX  MULTSZ-1 *XSAV1=MULTSZ(L)
9109:DO 02      298      BNE  SKIP15

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910B:C6 09      299      DEC  XSAV2
910D:86 08      300 SKIP15 STX  XSAV1

910F:A5 EF      302      LDA  ZGNCD+4  *1ST CHN SMPG PREP
9111:8D DO CO   303      STA  AI13

9114:A9 00      305      LDA  #$00    *CHN VLTG CLEAR LOOP
9116:A2 06      306      LDX  #$06
9118:95 19      307 LOOP17 STA  CHN1V-1,X
911A:95 F9      308      STA  CHN3V-1,X
911C:CA        309      DEX
911D:DO F9      310      BNE  LOOP17

911F:A6 19      312 LOOP18 LDX  ZNMCHN
9121:A5 EF      313      LDA  ZGNCD+4  *1ST CHN ACT SMPG & ADD
9123:8D DO CO   314      STA  AI13
9126:20 OA 95   315      JSR  DELAY13
9129:18        316      CLC          *1ST CHN SMPG & ADD
912A:AC D1 CO   317      LDY  AI13+1
912D:AD DO CO   318      LDA  AI13
9130:65 1A      319      ADC  CHN1V
9132:85 1A      320      STA  CHN1V
9134:A5 EE      321      LDA  ZGNCD+3  *2ND CHN SMPG PREPARATION
9136:8D DO CO   322      STA  AI13    *
9139:98        323      TYA
913A:29 OF      324      AND  #$0F
913C:65 1B      325      ADC  CHN1V+1
913E:85 1B      326      STA  CHN1V+1
9140:90 O2      327      BCC  SKIP16
9142:E6 1C      328      INC  CHN1V+2
9144:CA        329 SKIP16 DEX
9145:EA        330      NOP
9146:FO 6C      331      BEQ  NEXT11

9148:A5 EE      333      LDA  ZGNCD+3  *2ND CHN ACT SMPG
914A:8D DO CO   334      STA  AI13
914D:20 OA 95   335      JSR  DELAY13
9150:18        336      CLC          *2ND CHN SMPG & ADD
9151:AC D1 CO   337      LDY  AI13+1
9154:AD DO CO   338      LDA  AI13
9157:65 1D      339      ADC  CHN2V
9159:85 1D      340      STA  CHN2V
915B:A5 ED      341      LDA  ZGNCD+2  *3RD CHN SMPG PREP
915D:8D DO CO   342      STA  AI13    *
9160:98        343      TYA

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9161:29	OF	344	AND	#\$OF	
9163:65	1E	345	ADC	CHN2V+1	
9165:85	1E	346	STA	CHN2V+1	
9167:90	O2	347	BCC	SKIP17	
9169:E6	1F	348	INC	CHN2V+2	
916B:CA		349	SKIP17	DEX	
916C:EA		350		NOP	
916D:FO	45	351	BEQ	NEXT11	
916F:A5	ED	353	LDA	ZGNCD+2	*3RD CHN ACT SMPG
9171:8D	DO CO	354	STA	AI13	
9174:20	OA 95	355	JSR	DELAY13	
9177:18		356	CLC		*3RD CHN SMPG & ADD
9178:AC	D1 CO	357	LDY	AI13+1	
917B:AD	DO CO	358	LDA	AI13	
917E:65	FA	359	ADC	CHN3V	
9180:85	FA	360	STA	CHN3V	
9182:A5	EC	361	LDA	ZGNCD+1	*4TH CHN SMPG PREP
9184:8D	DO CO	362	STA	AI13	*
9187:98		363	TYA		
9188:29	OF	364	AND	#\$OF	
918A:65	FB	365	ADC	CHN3V+1	
918C:85	FB	366	STA	CHN3V+1	
918E:90	O2	367	BCC	SKIP18	
9190:E6	FC	368	INC	CHN3V+2	
9192:CA		369	SKIP18	DEX	
9193:EA		370		NOP	
9194:FO	1E	371	BEQ	NEXT11	
9196:A5	EC	373	LDA	ZGNCD+1	*4TH CHN ACT SMPG
9198:8D	DO CO	374	STA	AI13	
919B:20	OA 95	375	JSR	DELAY13	
919E:18		376	CLC		*4TH CHN SMPG & ADD
919F:AC	D1 CO	377	LDY	AI13+1	
91A2:AD	DO CO	378	LDA	AI13	
91A5:65	FD	379	ADC	CHN4V	
91A7:85	FD	380	STA	CHN4V	
91A9:98		381	TYA		
91AA:29	OF	382	AND	#\$OF	
91AC:65	FE	383	ADC	CHN4V+1	
91AE:85	FE	384	STA	CHN4V+1	
91B0:90	O2	385	BCC	NEXT11	
91B2:E6	FF	386	INC	CHN4V+2	
91B4:A5	EF	388	NEXT11	LDA	ZGNCD+4 *1ST CHN SMPG PREP



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91B6:8D DO CO 389 STA AI13 *
91B9:A5 O6 390 LDA ZLOC *5 CYCLES DELAY
91BB:EA 391 NOP

```

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91BC:C6 O8 393 DEC XSAV1
91BE:DO O5 394 BNE SKIP29
91CO:C6 O9 395 DEC XSAV2
91C2:10 O1 396 BPL SKIP29
91C4:60 397 RTS
91C5:4C 1F 91 398 SKIP29 JMP LOOP18

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91C8: 400 *SINGLE DATA GETTING SUBROUTINE

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91C8:A6 19 402 SNGLDAT LDX ZNMCHN *X=ZNMCHN
91CA:B5 EB 403 LOOP33 LDA ZGNCD,X *EACH CHN SMPG PREP
91CC:8D DO CO 404 STA AI13
91CF:20 O7 95 405 JSR DELAY19

```

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91D2:B5 EB 407 LDA ZGNCD,X *EACH CHN ACT SMPG
91D4:8D DO CO 408 STA AI13
91D7:C8 409 INY *10 CYCLE DELAY & ARRAY PTR UPDATE
91D8:FO O4 410 BEQ SKIP32 *
91DA:84 F9 411 STY YSAV1 *DUMMY
91DC:DO O2 412 BNE SKIP33 *
91DE:E6 O7 413 SKIP32 INC ZPAGE *
91EO:18 414 SKIP33 CLC *9 CYCLES DELAY & H-BY SAVE
91E1:AD D1 CO 415 LDA AI13+1 *
91E4:48 416 PHA *

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91E5:AD DO CO 418 LDA AI13 *8LSB ADD
91E8:71 O6 419 ADC (ZLOC),Y
91EA:91 O6 420 STA (ZLOC),Y
91EC:C8 421 INY
91ED:68 422 PLA *4MSB ADD
91EE:29 OF 423 AND #$0F
91FO:71 O6 424 ADC (ZLOC),Y
91F2:91 O6 425 STA (ZLOC),Y
91F4:C8 426 INY
91F5:A9 OO 427 LDA #$00
91F7:71 O6 428 ADC (ZLOC),Y

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91F9:91 06      429      STA  (ZLOC),Y
91FB:C8          430      INY

91FC:CA          432      DEX
91FD:DO CB      433      BNE  LOOP33

91FF:38          435      SEC          *INTERVAL DELAY(MIN 15 MICRO-SEC)
9200:A5 09      436      LDA  XSAV2   *INT
9202:A6 08      437      LDX  XSAV1   *INT
9204:FO 03      438      BEQ  SKIP34  *INT
9206:CA          439  LOOP34 DEX          *INT
9207:DO FD      440      BNE  LOOP34  *INT
9209:E9 01      441  SKIP34 SBC  #$01   *INT
920B:10 F9      442      BPL  LOOP34  *INT

920D:C6 FE      444      DEC  CHN4V+1 *LOOPING TIME COUNT
920F:DO B7      445      BNE  SNGLDAT
9211:C6 FF      446      DEC  CHN4V+2
9213:10 B3      447      BPL  SNGLDAT
9215:60          448      RTS

```

9216: 450 \*ARRAY CLEAR SUBROUTINE

```

9216:A9 00      452  PARYCLR LDA  #$00
9218:AO 14      453      LDY  #$14   *GTOT VAR CLEAR
921A:99 10 8F   454  LOOP16 STA  GTOT1-1,Y
921D:99 45 8F   455      STA  STOT1-1,Y
9220:88          456      DEY
9221:DO F7      457      BNE  LOOP16

9223:8D 2E 8F   459      STA  EXCMLN-1 *EXC(M/S)LN CLEAR
9226:8D 2F 8F   460      STA  EXCMLN
9229:8D 3E 8F   461      STA  EXCSLN-1
922C:8D 3F 8F   462      STA  EXCSLN
922F:60          463      RTS

9230:20 F3 94   465  DARYCLR JSR  PTRSET

9233:AE 01 8F   467      LDX  ARYSZ   *XSAV1=ARYSZ(H)
9236:86 08      468      STX  XSAV1
9238:AE 00 8F   469      LDX  ARYSZ-1 *X=ARYSZ(L)

```

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923B:DO O2      470      BNE  SKIP11
923D:C6 O8      471      DEC  XSAV1

923F:A9 O0      473  SKIP11 LDA  #$00
9241:C8          474  LOOP14 INY                *DATA ARRAY CLEAR
9242:DO O2      475      BNE  SKIP12
9244:E6 O7      476      INC  ZPAGE
9246:91 O6      477  SKIP12 STA  (ZLOC),Y
9248:CA          478      DEX
9249:DO F6      479      BNE  LOOP14
924B:C6 O8      480      DEC  XSAV1
924D:10 F2      481      BPL  LOOP14
924F:60          482      RTS

9250:A9 O0      484  TCARYCL LDA  #$00
9252:A2 O6      485      LDX  #$06          *CLOCK VAR CLEAR
9254:9D 2F 8F   486  LOOP15 STA  TIMEIN-1,X
9257:9D 3F 8F   487      STA  LAPTIM1-1,X
925A:9D 35 8F   488      STA  LAPTIM2-1,X
925D:95 19      489      STA  CHN1V-1,X *CHN VLTG CLEAR
925F:95 F9      490      STA  CHN3V-1,X
9261:CA          491      DEX
9262:DO FO      492      BNE  LOOP15
9264:60          493      RTS

9265:           495  *MAIN SW WAITING SUBROUTINE

9265:20 6E 92   497  MSWAIT JSR  MSWCHK
9268:FO FB      498      BEQ  MSWAIT
926A:20 FE 94   499      JSR  BELL
926D:60          500      RTS

926E:           502  *MAIN SW CHECKING SUBR

926E:A5 EB      504  MSWCHK LDA  ZGNCD
9270:8D DO CO   505      STA  AI13
9273:20 OB 95   506      JSR  DELAYO
9276:AD D1 CO   507      LDA  AI13+1
9279:29 OF      508      AND  #$0F
927B:60          509      RTS

927C:           511  *SCREEN CLEAR SUBROUTINE

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927C:A9 00      513  SCRNCLR LDA  #$00
927E:85 22      514          STA  $22
9280:20 58 FC    515          JSR  $FC58
9283:60          516          RTS

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9284:          518 *HEADLINE DISP & CHOICE SUBR

```

```

9284:20 13 93    520  HDLNCHO JSR  CAPTDISP
9287:20 35 FD    521          JSR  $FD35 *GET A KEY F. KEYBOARD
928A:8D 25 8F    522          STA  CHOICE
928D:60          523          RTS

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928E:          525 *INITIAL LOOPING LMT SET SUBR

```

```

928E:B9 0B 8F    527  LPLMTST LDA  LNMULT,Y *ZLPNUM=(LNMULT,Y)(H)
9291:85 CF      528          STA  ZLPNUM
9293:B9 0A 8F    529          LDA  LNMULT-1,Y *ZLPNUM-1=(LNMULT-1,Y)(L)
9296:DO 02      530          BNE  SKIP14
9298:C6 CF      531          DEC  ZLPNUM *DECREASE ZLPNUM BY 1
929A:85 CE      532  SKIP14 STA  ZLPNUM-1
929C:60          533          RTS

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929D:          535 *ELAPSED TIME CALCULATION SUBR

```

```

929D:AE C4 CO    537  STOPTM LDX  CLOCK+4 *10 TO 1 MIL-SEC(BCD FORMAT)
92A0:AD C3 CO    538          LDA  CLOCK+3 *2^3--2^0 SEC & 100 MIL-SEC(BCD)
92A3:AC C2 CO    539          LDY  CLOCK+2 *2^11--2^4 SEC
92A6:8E 45 8F    540          STX  LAPTIM1+5 *LAPTIM1+5=10 TO 1 MIL-SEC(BCD FMT)

92A9:D8          542          CLD
92AA:AA          543          TAX
92AB:29 OF      544          AND  #$OF
92AD:8D 44 8F    545          STA  LAPTIM1+4 *LAPTIM1+4=100 MIL-SEC(BCD FMT)

92B0:8A          547          TXA
92B1:29 FO      548          AND  #$FO
92B3:38          549          SEC
92B4:ED 33 8F    550          SBC  TIMEIN+3

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92B7:08      551      PHP          *P-REGISTER SAVE
92B8:4A      552      LSR A        *DIVIDE A-REG BY 16
92B9:4A      553      LSR A
92BA:4A      554      LSR A
92BB:4A      555      LSR A
92BC:8D 43 8F 556      STA LAPTIM1+3 *LAPTIM1+3=2^3--2^0 SEC

92BF:98      558      TYA
92CO:28      559      PLP          *P-REGISTER RESTORE
92C1:ED 32 8F 560      SBC TIMEIN+2
92C4:8D 42 8F 561      STA LAPTIM1+2 *LAPTIM1+2=2^11--2^4 SEC
92C7:60      562      RTS

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92C8:          564 *LAPTIME MOVE SUBR

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```

92C8:A2 06      566 LAPTMV LDX #$06
92CA:BD 3F 8F 567 LOOP71 LDA LAPTIM1-1,X
92CD:9D 35 8F 568      STA LAPTIM2-1,X
92DO:CA          569      DEX
92D1:DO F7      570      BNE LOOP71
92D3:60          571      RTS

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92D4:          573 *INITIAL TIME SET SUBROUTINE

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92D4:AD C6 CO 575 INITIM LDA CLOCK+6 *CLOCK STOP
92D7:AD C5 CO 576      LDA CLOCK+5 *CLOCK RESTART

92DA:AD C3 CO 578      LDA CLOCK+3 *2^3--2^0 SEC READ
92DD:AE C2 CO 579      LDX CLOCK+2 *2^11--2^4 SEC READ
92EO:8D 33 8F 580      STA TIMEIN+3
92E3:8E 32 8F 581      STX TIMEIN+2
92E6:60          582      RTS

```

```

92E7:          584 *TIME STEP CHECK SUBR:500 MIL-SEC

```

```

92E7:18          586 TMSTEP1 CLC
92E8:F8          587      SED          *SET DECIMAL FLAG
92E9:AD 3C 8F 588      LDA REFTIM1 *REFTIM1=REFTIM1+INTERVAL
92EC:69 05      589      ADC #$05    *INTERVAL=500 MIL-SEC
92EE:29 0F      590      AND #$0F
92FO:8D 3C 8F 591      STA REFTIM1
92F3:D8          592      CLD

```

```

92F4:AD C3 CO 593 LOOP19 LDA CLOCK+3 *2^3--2^0 & 100 MIL-SEC(BCD) READ
92F7:29 OF 594 AND #$0F
92F9:CD 3C 8F 595 CMP REFTIM1
92FC:DO F6 596 BNE LOOP19
92FE:60 597 RTS

```

```

92FF:18 599 TMSTEP2 CLC *TIME STEP=1 SEC
9300:AD 3D 8F 600 LDA REFTIM2
9303:69 10 601 ADC #$10
9305:8D 3D 8F 602 STA REFTIM2
9308:AD C3 CO 603 LOOP51 LDA CLOCK+3 *SEC &100 MIL-SEC
930B:29 FO 604 AND #$FO
930D:CD 3D 8F 605 CMP REFTIM2
9310:DO F6 606 BNE LOOP51
9312:60 607 RTS

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9313: 609 *CAPTION DISPLAY SUBROUTINE

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```

9313:BD OC 95 611 CAPTDISP LDA CAPTN1,X
9316:FO O6 612 BEQ RETURN
9318:20 ED FD 613 JSR $FDED
931B:E8 614 INX
931C:DO F5 615 BNE CAPTDISP
931E:60 616 RETURN RTS

```

```

931F: 618 *MULTIPLE DATA STORE SUBROUTINE

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931F:A4 F9 620 STORE LDY YSAV1 *Y=YSAV1
9321:A6 19 621 LDX ZNMCHN *XSAV1=ZNMCHN
9323:86 O8 622 STX XSAV1

9325:A2 O0 624 LDX #$00 *CHN1V SAVE
9327:20 42 93 625 JSR CHNVSTO
932A:FO 13 626 BEQ NEXT12
932C:A2 O3 627 LDX #$03 *CHN2V SAVE
932E:20 42 93 628 JSR CHNVSTO
9331:FO OC 629 BEQ NEXT12
9333:A2 EO 630 LDX #$EO *CHN3V STORE
9335:20 42 93 631 JSR CHNVSTO

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9338:FO 05      632      BEQ  NEXT12
933A:A2 E3      633      LDX  #$E3      *CHN4V STORE
933C:20 42 93   634      JSR  CHNVSTO

933F:84 F9      636 NEXT12  STY  YSAV1
9341:60          637      RTS

9342:C8          639 CHNVSTO  INY
9343:DO 02      640      BNE  SKIP19
9345:E6 07      641      INC  ZPAGE
9347:B5 1A      642 SKIP19  LDA  CHN1V,X
9349:91 06      643      STA  (ZLOC),Y
934B:C8          644      INY
934C:B5 1B      645      LDA  CHN1V+1,X
934E:91 06      646      STA  (ZLOC),Y
9350:C8          647      INY
9351:B5 1C      648      LDA  CHN1V+2,X
9353:91 06      649      STA  (ZLOC),Y
9355:C8          650      INY
9356:C6 08      651      DEC  XSAV1
9358:60          652      RTS

9359:           654 *DIRECT ADDITION OF CHNV INTO G/S-TOT

9359:A6 19      656 ADDTN  LDX  ZNMCHN
935B:86 08      657      STX  XSAV1

935D:A6 09      659      LDX  XSAV2      *GTOT1=$00 (XSAV2) STOT1=$35
935F:AO 00      660      LDY  #$00      *CHN1V ADDITION
9361:20 89 93   661      JSR  CHNVADD
9364:FO 1C      662      BEQ  NEXT13

9366:20 83 93   664      JSR  INCX5
9369:AO 03      665      LDY  #$03      *CHN2V ADDITION
936B:20 89 93   666      JSR  CHNVADD
936E:FO 12      667      BEQ  NEXT13

9370:20 83 93   669      JSR  INCX5
9373:AO EO      670      LDY  #$EO      *CHN3V ADDITION
9375:20 89 93   671      JSR  CHNVADD
9378:FO 08      672      BEQ  NEXT13

937A:20 83 93   674      JSR  INCX5

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937D:A0 E3 675 LDY #E3 \*CHN4V ADDITION  
 937F:20 89 93 676 JSR CHNVADD

9382:60 678 NEXT13 RTS

9383:18 680 INCX5 CLC  
 9384:8A 681 TXA  
 9385:69 05 682 ADC #05  
 9387:AA 683 TAX  
 9388:60 684 RTS

9389:18 686 CHNVADD CLC  
 938A:B9 1A 00 687 LDA CHN1V,Y  
 938D:7D 11 8F 688 ADC GTOT1,X  
 9390:9D 11 8F 689 STA GTOT1,X  
 9393:B9 1B 00 690 LDA CHN1V+1,Y  
 9396:7D 12 8F 691 ADC GTOT1+1,X  
 9399:9D 12 8F 692 STA GTOT1+1,X  
 939C:B9 1C 00 693 LDA CHN1V+2,Y  
 939F:7D 13 8F 694 ADC GTOT1+2,X  
 93A2:9D 13 8F 695 STA GTOT1+2,X  
 93A5:90 08 696 BCC SKIP23  
 93A7:FE 14 8F 697 INC GTOT1+3,X  
 93AA:DO 03 698 BNE SKIP23  
 93AC:FE 15 8F 699 INC GTOT1+4,X  
 93AF:C6 08 700 SKIP23 DEC XSAV1  
 93B1:60 701 RTS

93B2: 703 \*LOOPING NUMBER DISPLAY SUBR

93B2:20 58 FC 705 LNDISP JSR \$FC58  
 93B5:E6 FC 706 INC CHN3V+2  
 93B7:DO 02 707 BNE SKIP42  
 93B9:E6 FD 708 INC CHN4V  
 93BB:A5 FD 709 SKIP42 LDA CHN4V \*H-BYTE  
 93BD:20 DA FD 710 JSR \$FD0A  
 93C0:A5 FC 711 LDA CHN3V+2 \*L-BYTE  
 93C2:20 DA FD 712 JSR \$FD0A  
 93C5:60 713 RTS



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```

93C6:          715 *ARRAY DISPLAY SUBROUTINE

93C6:A2 09      717 ARYDISP LDX  #$09      *CAPTION 2 DISPLAY
93C8:20 13 93   718          JSR  CAPTDISP
93CB:20 F3 94   719          JSR  PTRSET

93CE:A5 19      721          LDA  ZNMCHN
93D0:8D 2C 8F   722          STA  DLN1
93D3:A9 04      723          LDA  #$04
93D5:8D 2D 8F   724          STA  DLN2
93D8:A9 02      725          LDA  #$02      *VTAB=02
93DA:85 22      726          STA  $22

93DC:20 35 FD   728 DISPASK JSR  $FD35
93DF:C9 D9      729          CMP  #'Y'
93E1:DO 06      730          BNE  SKIP51
93E3:20 C3 94   731          JSR  MEMDISP
93E6:4C DC 93   732          JMP  DISPASK

93E9:A9 00      734 SKIP51  LDA  #$00      *VTAB=00
93EB:85 22      735          STA  $22
93ED:60         736          RTS

93EE:          738 *GTOT/STOT & TOTAL LAPTME DISP & BELL

93EE:A4 19      740 GTOTLAPT LDY ZNMCHN
93FO:84 09      741          STY  XSAV2
93F2:A0 05      742 LOOP28  LDY  #$05
93F4:BD 11 8F   743 LOOP27  LDA  GTOT1,X
93F7:20 DA FD   744          JSR  $FDDA
93FA:E8         745          INX
93FB:88         746          DEY
93FC:DO F6      747          BNE  LOOP27
93FE:A9 8D      748          LDA  #$8D
9400:20 ED FD   749          JSR  $FDED
9403:C6 09      750          DEC  XSAV2
9405:DO EB      751          BNE  LOOP28

9407:20 2A 94   753          JSR  LAPTDISP
940A:20 FE 94   754          JSR  BELL
940D:60         755          RTS

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940E:          757 *CHN VLTG & LAP TIME DISPLAY SUBROUTINE

940E:A4 19      759 DISPLY LDY ZNMCHN

9410:A2 00      761          LDX #$00      *CHN1V DISP
9412:20 48 94  762          JSR CHNVDISP
9415:FO 13      763          BEQ LAPTDISP
9417:A2 03      764          LDX #$03      *CHN2V DISP
9419:20 48 94  765          JSR CHNVDISP
941C:FO 0C      766          BEQ LAPTDISP
941E:A2 E0      767          LDX #$E0      *CHN3V DISP
9420:20 48 94  768          JSR CHNVDISP
9423:FO 05      769          BEQ LAPTDISP
9425:A2 E3      770          LDX #$E3      *CHN4V DISP
9427:20 48 94  771          JSR CHNVDISP

942A:AD 42 8F  773 LAPTDISP LDA LAPTIM1+2 *2^11--2^4 SEC
942D:20 DA FD  774          JSR $FDDA
9430:AD 43 8F  775          LDA LAPTIM1+3 *2^3--0 SEC
9433:20 DA FD  776          JSR $FDDA
9436:AD 44 8F  777          LDA LAPTIM1+4 *100 MIL SEC
9439:20 DA FD  778          JSR $FDDA
943C:AD 45 8F  779          LDA LAPTIM1+5 *10--1 MIL SEC
943F:20 DA FD  780          JSR $FDDA
9442:A9 8D      781          LDA #$8D
9444:20 ED FD  782          JSR $FDED
9447:60          783          RTS

9448:B5 1C      785 CHNVDISP LDA CHN1V+2,X
944A:20 DA FD  786          JSR $FDDA
944D:B5 1B      787          LDA CHN1V+1,X
944F:20 DA FD  788          JSR $FDDA
9452:B5 1A      789          LDA CHN1V,X
9454:20 DA FD  790          JSR $FDDA
9457:A9 AO      791          LDA #$AO
9459:20 ED FD  792          JSR $FDED
945C:88          793          DEY
945D:60          794          RTS

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945E:          796 *FINAL EXEC'TED LOOP NO CALC

945E:A5 CF      798 EXLPNM  LDA  ZLPNUM
9460:10 04      799          BPL  SKIP91  *IF ZLPNUM(H)=>$00 THEN BR
9462:E6 CF      800          INC  ZLPNUM
9464:10 02      801          BPL  SKIP98  *UNCOND BR(NOW ZLPNUM(H)=$00)
9466:C6 CE      802 SKIP91  DEC  ZLPNUM-1
9468:38         803 SKIP98  SEC
9469:B9 0A 8F   804          LDA  LNMULT-1,Y
946C:E5 CE      805          SBC  ZLPNUM-1
946E:9D 2E 8F   806          STA  EXCMLN-1,X
9471:B9 0B 8F   807          LDA  LNMULT,Y
9474:E5 CF      808          SBC  ZLPNUM
9476:9D 2F 8F   809          STA  EXCMLN,X
9479:60         810          RTS

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947A:          812 *MULT/SNGL DATA ADD SUBROUTINE

947A:20 8E 92   814 MULTADD JSR  LPLMTST  *ZLPNUM=EXCMLN OR LNSNGL
947D:20 F3 94   815          JSR  PTRSET

9480:A6 09      817 LOOP98  LDX  XSAV2
9482:A5 19      818          LDA  ZNMCHN
9484:85 08      819          STA  XSAV1
9486:C8         820 LOOP99  INY
9487:DO 02      821          BNE  SKIP97
9489:E6 07      822          INC  ZPAGE
948B:18         823 SKIP97  CLC
948C:B1 06      824          LDA  (ZLOC),Y  *1ST BYTE
948E:7D 11 8F   825          ADC  GTOT1,X
9491:9D 11 8F   826          STA  GTOT1,X
9494:C8         827          INY
9495:B1 06      828          LDA  (ZLOC),Y  *2ND BYTE
9497:7D 12 8F   829          ADC  GTOT1+1,X
949A:9D 12 8F   830          STA  GTOT1+1,X
949D:C8         831          INY
949E:B1 06      832          LDA  (ZLOC),Y  *3RD BYTE
94A0:7D 13 8F   833          ADC  GTOT1+2,X
94A3:9D 13 8F   834          STA  GTOT1+2,X
94A6:C8         835          INY
94A7:90 08      836          BCC  SKIP95
94A9:FE 14 8F   837          INC  GTOT1+3,X
94AC:DO 03      838          BNE  SKIP95

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94AE:FE 15 8F 839 INC GTOT1+4,X
94B1:8A 840 SKIP95 TXA
94B2:18 841 CLC
94B3:69 05 842 ADC #$05
94B5:AA 843 TAX
94B6:C6 08 844 DEC XSAV1
94B8:DO CC 845 BNE LOOP99

94BA:C6 CE 847 DEC ZLPNUM-1
94BC:DO C2 848 BNE LOOP98
94BE:C6 CF 849 DEC ZLPNUM
94CO:10 BE 850 BPL LOOP98
94C2:60 851 RTS

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94C3: 853 *DATA ARRAY DISPLAY SUBROUTINE; 20 LINES AT A TIME
94C3: 854 *TO CONTINUE TO DISPLAY, TYPE "Y"

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94C3:A4 F9 856 MEMDISP LDY YSAV1
94C5:A9 14 857 LDA #$14 *XSAV1=20
94C7:85 08 858 STA XSAV1
94C9:AE 2C 8F 859 LOOP25 LDX DLN1 *XSAV2=DLN1
94CC:86 09 860 STX XSAV2
94CE:AE 2D 8F 861 LOOP24 LDX DLN2 *X=DLN2

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94D1:C8 863 LOOP23 INY
94D2:DO 02 864 BNE SKIP26
94D4:E6 07 865 INC ZPAGE
94D6:B1 06 866 SKIP26 LDA (ZLOC),Y
94D8:20 DA FD 867 JSR $FDDA
94DB:CA 868 DEX
94DC:DO F3 869 BNE LOOP23
94DE:A9 A0 870 LDA #$A0
94EO:20 ED FD 871 JSR $FDED

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94E3:C6 09 873 DEC XSAV2
94E5:DO E7 874 BNE LOOP24
94E7:A9 8D 875 LDA #$8D
94E9:20 ED FD 876 JSR $FDED

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94EC:C6 08 878 DEC XSAV1
94EE:DO D9 879 BNE LOOP25

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94F0:84 F9 880 STY YSAV1  
 94F2:60 881 RTS

94F3: 883 \*POINTER SET SUBROUTINE

94F3:AC 03 8F 885 PTRSET LDY STALOC \*ZPAGE=STALOC(H)  
 94F6:84 07 886 STY ZPAGE  
 94F8:AC 02 8F 887 LDY STALOC-1 \*YSAV1=STALOC(L)  
 94FB:84 F9 888 STY YSAV1  
 94FD:60 889 RTS

94FE: 891 \*RING-BELL SUBROUTINE

94FE:A9 87 893 BELL LDA #\$87  
 9500:20 ED FD 894 JSR \$FDED  
 9503:20 ED FD 895 JSR \$FDED  
 9506:60 896 RTS

9507: 898 \*TIME DELAY SUBROUTINE

9507:A5 06 900 DELAY19 LDA ZLOC  
 9509:EA 901 NOP  
 950A:EA 902 DELAY13 NOP  
 950B:60 903 DELAYO RTS

950C: 905 \*DATA AND CONSTANTS

950C:C2 C1 D3 907 CAPTN1 ASC 'BASE ' \*F-D;\$4F,F-N;\$4E  
 950F:C5 AO  
 9511:4F 4E AO 908 DFB \$4F,\$4E,\$AO,\$BA  
 9514:BA  
 9515:C4 BF AO 909 CAPTN2 ASC 'D? Y/N'  
 9518:D9 AF CE  
 951B:8D 00 910 DFB \$8D,\$00  
 951D:C1 C3 D4 911 CAPTN3 ASC 'ACTV '  
 9520:D6 AO  
 9522:4F 4E AO 912 DFB \$4F,\$4E,\$AO,\$BA  
 9525:BA  
 9526:C4 BF AO 913 ASC 'D? Y/N'  
 9529:D9 AF CE  
 952C:8D 00 914 DFB \$8D,\$00

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952E:C2 C1 D3 915 CAPTN4 ASC 'BASE ' *F-F;$46
9531:C5 AO
9533:4F 46 46 916 DFB $4F,$46,$46,$8D,$00
9536:8D 00
9538:C1 C3 D4 917 CAPTN5 ASC 'ACTV '
953B:D6 AO
953D:4F 46 46 918 DFB $4F,$46,$46,$8D,$00
9540:8D 00
9542:D3 CE C7 919 CAPTN6 ASC 'SNGL '
9545:CC AO
9547:4F 4E AO 920 DFB $4F,$4E,$AO,$BA
954A:BA
954B:C4 BF AO 921 ASC 'D? Y/N'
954E:D9 AF CE
9551:8D 00 922 DFB $8D,$00
9553:D3 CE C7 923 CAPTN7 ASC 'SNGL '
9556:CC AO
9558:4F 46 46 924 DFB $4F,$46,$46,$8D,$00
955B:8D 00
955D: 925 *PROGRAMMED BY TAEWOOK KANG, MAY 5,1983
955D: 926 *REVISED BY TAEWOOK KANG, AUG.23,1983
955D: 927 *RECONSTRUCTED BY TAEWOOK KANG, SEPT. 1,1983
955D: 928 *REMODIFIED FOR HOT W/F ANNEMO BY TAEWOOK KANG, FEB. 20,1984

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\*\*\* SUCCESSFUL ASSEMBLY: NO ERRORS

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?8F05 ACTSTL          ?9359 ADDTN
  CODO AI13           93C6 ARYDISP
  8F01 ARYSZ          94FE BELL
  8FA3 BSVLTG         9313 CAPTDISP
  950C CAPTN1         ?9515 CAPTN2
  ?951D CAPTN3        ?952E CAPTN4
  ?9538 CAPTN5        ?9542 CAPTN6
  ?9553 CAPTN7        1A CHN1V
    1D CHN2V          FA CHN3V
    FD CHN4V          9389 CHNVADD
  9448 CHNVDISP       9342 CHNVSTO
  8F25 CHOICE         COCO CLOCK
  9230 DARYCLR        950B DELAYO
  950A DELAY13        9507 DELAY19
  93DC DISPASK        940E DISPLY
  8F2C DLN1           8F2D DLN2
  ?8F07 ENDLOC        8F2F EXCMLN
  8F3F EXCSLN         9042 EXIT14
  904F EXIT15         90D5 EXIT22

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90C5	EXIT33	8FD6	EXIT74
8FE3	EXIT75	945E	EXLPNM
?900F	GETMULT	9078	GETSNGL
8F26	GNCB	8F11	GTOT1
?8F16	GTOT2	?8F1B	GTOT3
?8F20	GTOT4	93EE	GTOTLAPT
9284	HDLNCHO	9383	INCX5
92D4	INITIM	8FOF	INTVL
942A	LAPTDISP	8F40	LAPTIM1
8F36	LAPTIM2	92C8	LPTMV
93B2	LNDISP	8FOB	LNMULT
8FOD	LNSNGL	9016	LOOP12
9241	LOOP14	9254	LOOP15
921A	LOOP16	9118	LOOP17
911F	LOOP18	92F4	LOOP19
902F	LOOP20	94D1	LOOP23
94CE	LOOP24	94C9	LOOP25
93F4	LOOP27	93F2	LOOP28
9084	LOOP32	91CA	LOOP33
9206	LOOP34	909D	LOOP41
9308	LOOP51	8FC3	LOOP70
92CA	LOOP71	8FAA	LOOP72
9480	LOOP98	9486	LOOP99
928E	LPLMTST	94C3	MEMDISP
9265	MSWAIT	926E	MSWCHK
947A	MULTADD	9101	MULTDAT
8F09	MULTSZ	91B4	NEXT11
933F	NEXT12	9382	NEXT13
8F10	NMCHN	8F80	ORIGIN
9216	PARYCLR	94F3	PTRSET
8F3C	REFTIM1	8F3D	REFTIM2
931E	RETURN	927C	SCRNCLR
923F	SKIP11	9246	SKIP12
929A	SKIP14	910D	SKIP15
9144	SKIP16	916B	SKIP17
9192	SKIP18	9347	SKIP19
93AF	SKIP23	94D6	SKIP26
91C5	SKIP29	90AC	SKIP31
91DE	SKIP32	91E0	SKIP33
9209	SKIP34	93BB	SKIP42
93E9	SKIP51	8FA0	SKIP77
9466	SKIP91	94B1	SKIP95
948B	SKIP97	9468	SKIP98
91CB	SNGLDAT	8F03	STALOC
929D	STOPTM	931F	STORE

8F46 STOT1	?8F4B STOT2
?8F50 STOT3	?8F55 STOT4
9250 TCARYCL	8F30 TIMEIN
?92E7 TMSTEP1	92FF TMSTEP2
08 XSAV1	09 XSAV2
F9 YSAV1	EB ZGNCD
06 ZLOC	CF ZLPNUM
19 ZNMCHN	07 ZPAGE
06 ZLOC	07 ZPAGE
08 XSAV1	09 XSAV2
19 ZNMCHN	1A CHN1V
1D CHN2V	CF ZLPNUM
EB ZGNCD	F9 YSAV1
FA CHN3V	FD CHN4V
8F01 ARYSZ	8F03 STALOC
?8F05 ACTSTL	?8F07 ENDLOC
8F09 MULTSZ	8F0B LNMULT
8F0D LNSNGL	8F0F INTVL
8F10 NMCHN	8F11 GTOT1
?8F16 GTOT2	?8F1B GTOT3
?8F20 GTOT4	8F25 CHOICE
8F26 GNCD	8F2C DLN1
8F2D DLN2	8F2F EXCMLN
8F30 TIMEIN	8F36 LAPTIM2
8F3C REFTIM1	8F3D REFTIM2
8F3F EXCSLN	8F40 LAPTIM1
8F46 STOT1	?8F4B STOT2
?8F50 STOT3	?8F55 STOT4
8F80 ORIGIN	8FA0 SKIP77
8FA3 BSVLTG	8FAA LOOP72
8FC3 LOOP70	8FD6 EXIT74
8FE3 EXIT75	?900F GETMULT
9016 LOOP12	902F LOOP20
9042 EXIT14	904F EXIT15
9078 GETSNGL	9084 LOOP32
909D LOOP41	90AC SKIP31
90C5 EXIT33	90D5 EXIT22
9101 MULTDAT	910D SKIP15
9118 LOOP17	911F LOOP18
9144 SKIP16	916B SKIP17
9192 SKIP18	91B4 NEXT11
91C5 SKIP29	91C8 SNGLDAT
91CA LOOP33	91DE SKIP32
91E0 SKIP33	9206 LOOP34



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9209 SKIP34	9216 PARYCLR
921A LOOP16	9230 DARYCLR
923F SKIP11	9241 LOOP14
9246 SKIP12	9250 TCARYCL
9254 LOOP15	9265 MSWAIT
926E MSWCHK	927C SCRNCCLR
9284 HDLNCHO	928E LPLMTST
929A SKIP14	929D STOPTM
92C8 LAPTMV	92CA LOOP71
92D4 INITIM	?92E7 TMSTEP1
92F4 LOOP19	92FF TMSTEP2
9308 LOOP51	9313 CAPTDISP
931E RETURN	931F STORE
933F NEXT12	9342 CHNVSTO
9347 SKIP19	?9359 ADDTN
9382 NEXT13	9383 INCX5
9389 CHNVADD	93AF SKIP23
93B2 LNDISP	93BB SKIP42
93C6 ARYDISP	93DC DISPASK
93E9 SKIP51	93EE GTOTLAPT
93F2 LOOP28	93F4 LOOP27
940E DISPLY	942A LAPTDISP
9448 CHNVDISP	945E EXLPNM
9466 SKIP91	9468 SKIP98
947A MULTADD	9480 LOOP98
9486 LOOP99	948B SKIP97
94B1 SKIP95	94C3 MEMDISP
94C9 LOOP25	94CE LOOP24
94D1 LOOP23	94D6 SKIP26
94F3 PTRSET	94FE BELL
9507 DELAY19	950A DELAY13
950B DELAYO	950C CAPTN1
?9515 CAPTN2	?951D CAPTN3
?952E CAPTN4	?9538 CAPTN5
?9542 CAPTN6	?9553 CAPTN7
COCO CLOCK	CODO AI13

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100 TEXT : HOME : FLASH : HTAB 7: PRINT "FULL CONV OF SYS-DATA": NORMAL
200 DIM A(9),B(9),C(19),D(9),E(9),G(9),GA(9)
210 GOSUB 9700: REM HIMEM SETTING
290 K = 0
300 ARYSZ = 36609 + K:STALOC = 36611 + K:ACTSTL = 36613 + K
310 ENLOC = 36615 + K:MULTSZ = 36617 + K:MLPNM = 36619 + K
320 SLPNM = 36621 + K:INERVL = 36623 + K:NMCHN = 36624 + K
325 GSTAL = 36625 + K:GNCD = 36646 + K:TYPE = 36651 + K
330 MEXCLN = 36655 + K:TIMEIN = 36656 + K:L2AP = 36662 + K
340 R1EF = 36668 + K:R2EF = 36669 + K:SEXCLN = 36671 + K
350 L1AP = 36672 + K:S1TAL = 36678 + K:CLSYS = 36730 + K
360 G1R = 36698 + K:XPNO = 36702 + K:XCRD = 36704 + K
370 YCRD = 36706 + K:ZCRD = 36708 + K:RTN = 36710 + K
380 CURR = 36712 + K:TEMP = 36714 + K:OHR = 36716 + K:SVLTG = 36718
390 XDTE = 36732 + K:PARA = 36608 + K:HOT1 = 36736 + K
500 D$ = CHR$(4):R$ = CHR$(13):H = 256
1000 POKE 34,21: HOME : PRINT : INPUT " DRIVE NO.? ";D%
1010 POKE 34,2: HOME : PRINT R$;D$;"CATALOG,D";D%
1020 INPUT " DATA ACQ TYPE =? (M/S) ";TY$
1030 INPUT " EXP. NO. = ";XP
1040 F$ = "S-" + STR$(XP) + "-" + TY$:G$ = F$ + "-TS"
1050 PRINT R$;D$;"BLOAD ";F$;",";D";D%: HOME : VTAB 2: INVERSE : PRINT F$;",";D";D%: NORMAL : POKE 34,3
1060 NC = PEEK (NMCHN):A% = 4:NP = 7
1070 GOSUB 6000: REM GAIN CODE CALC:(D(I)) ,(GA(I))
1080 IF TY$ > < "S" THEN GOTO 1100
1090 GOSUB 3000: GOTO 1110: REM SNGL CONV
1100 GOSUB 2000: REM MULT CONV
1110 GOSUB 4000
1120 INPUT " WANT TO CONV TS ? (Y/N) ";AN$
1130 IF LEFT$(AN$,1) > < "Y" THEN GOTO 1150
1140 PRINT R$;D$;"BLOAD ";G$;",";D";D%: VTAB 2: INVERSE : PRINT G$;",";D";D%: NORMAL : POKE 34,3: GOSUB 4500
1150 PRINT R$;D$;"RUN MENU,D1"
1160 END
2000 DS = PEEK (MULTSZ) * H + PEEK (MULTSZ - 1):DAS=DS:DBS=DS
2010 NBD = PEEK (SEXCLN) * H + PEEK (SEXCLN - 1):NAD = PEEK (MEXCLN) * H + PEEK (MEXCLN - 1)
2020 FOR I = 0 TO NC - 1
2030 A(I) = 0:B(I) = 0
2040 FOR K = 0 TO 4
2050 B(I) = B(I) + PEEK (S1TAL + 5 * I + K) * (H ^ K) / (DBS * NBD)
2060 A(I) = A(I) + PEEK (GSTAL + 5 * I + K) * (H ^ K) / (DAS * NAD)
2070 NEXT K
2080 B(I) = D(I) * B(I) / 4095 / GA(I):A(I) = D(I) * A(I) / 4095 / GA(I)
2090 NEXT I
2100 MSB = INT ( PEEK (L2AP + 5) / 16):MTB = PEEK (L2AP + 5) - MSB * 16
2110 TBM = PEEK (L2AP + 2) * 16 + PEEK (L2AP + 3) + PEEK (L2AP + 4) / 10 + MSB / 100 + MTB / 1000
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2120 MSA = INT ( PEEK (L1AP + 5) / 16):MTA = PEEK (L1AP + 5) - MSA * 16
2130 TAM = PEEK (L1AP + 2) * 16 + PEEK (L1AP + 3) + PEEK (L1AP + 4) / 10 + MSA / 100 + MTA / 1000
2140 RETURN : END
3000 DBS = PEEK (MULTSZ) * H + PEEK (MULTSZ - 1):DAS = PEEK (SEXCLN) * H + PEEK (SEXCLN - 1)
3005 NBD = PEEK (MEXCLN) * H + PEEK (MEXCLN - 1):NAD = PEEK (SLPNM) * H + PEEK (SLPNM - 1)
3010 FOR I = 0 TO NC - 1
3020 A(I) = 0:B(I) = 0
3030 FOR K = 0 TO 4
3040 B(I) = B(I) + PEEK (GSTAL + 5 * I + K) * (H ^ K) / (DBS * NBD)
3050 A(I) = A(I) + PEEK (S1TAL + 5 * I + K) * (H ^ K) / (DAS * NAD)
3060 NEXT K
3070 B(I) = D(I) * B(I) / 4095 / GA(I):A(I) = D(I) * A(I) / 4095 / GA(I)
3080 NEXT I
3090 MSB = INT ( PEEK (L1AP + 5) / 16):MTB = PEEK (L1AP + 5) - MSB * 16
3100 TBM = PEEK (L1AP + 2) * 16 + PEEK (L1AP + 3) + PEEK (L1AP + 4) / 10 + MSB / 100 + MTB / 1000
3110 MSA = INT ( PEEK (L2AP + 5) / 16):MTA = PEEK (L2AP + 5) - MSA * 16
3120 TAM = PEEK (L2AP + 2) * 16 + PEEK (L2AP + 3) + PEEK (L2AP + 4) / 10 + MSA / 100 + MTA / 1000
3130 RETURN : END
4000 POKE 34,3: HOME
4010 INPUT " WANT TO PRINT OUT? (Y/N) ";AN$: IF LEFT$ (AN$,1) > < "Y" THEN HOME : GOTO 4030
4020 PRINT R$;D$;"PR#1": POKE 1401,80: HOME
4030 C(O) = PEEK (XDTE + 3) + PEEK (XDTE + 2) * H + PEEK (XDTE + 1) * H * H + PEEK (XDTE) * H * H * H
4040 A$ = STR$ (C(O))
4050 FOR J = 0 TO NP - 1:C(J) = PEEK (XCRD + 2 * J) * H + PEEK (XCRD + 2 * J + 1): NEXT J
4060 C(O) = C(O) / 1E3:C(1) = C(1) / 1E3:C(2) = C(2) / 1E3:C(6) = C(6) / 100:C(5) = C(5) / 10
4070 FOR I = 1 TO 39: PRINT "*":: NEXT I: PRINT : PRINT
4080 PRINT "FILE NAME & EXP NO = ";TAB(20);F$;TAB(41);"EXPERIMENT DATE = ";LEFT$ (A$,4);"/";MID$ (A$,5,2);"/";RIGHT$ (A$,2)
4090 PRINT "(X/R)= ";C(O);" CM"; TAB( 21);"(Y/O)= ";C(1);" CM"; TAB( 41);"(Z)= ";C(2);" CM"
4100 PRINT "CURR= ";C(4);" AMP"; TAB( 21);"OP TEMP= ";C(5);" DEG"; TAB( 41);"O/H R= ";C(6)
4120 FOR J = 0 TO NC - 1
4130 PRINT "GNCD:CH-";J + 1;"=" ;G(J); TAB( 21);"REC GN:CH-";J + 1;"=" ;GA(J); TAB( 41);"SVLTG:CH-";J + 1;"=" ;E(J);" VOLT"
4140 PRINT "BASE VLTG: CH-";J + 1;"=" ;B(J);" VOLT"; TAB( 41);
4150 PRINT "ACTV VLTG: CH-";J + 1;"=" ;A(J);" VOLT"
4160 NEXT J
4170 PRINT "TOT SAMP TIM: BAS= ";TBM;" SEC"; TAB( 41);
4180 PRINT "TOT DAT PTS: BAS= ";DBS;"*";NBD;"=" ;DBS * NBD
4190 PRINT "TOT SAMP TIM: ACT= ";TAM;" SEC"; TAB( 41);
4200 PRINT "TOT DAT PTS: ACT= ";DAS;"*";NAD;"=" ;DAS * NAD
4210 PRINT : FOR I = 1 TO 39: PRINT "*":: NEXT I: PRINT
4220 PRINT R$;D$;"PR#0"
4230 RETURN : END
4500 POKE 34, PEEK (37): HOME : INPUT " WANT TO PRINT-OUT TS? (Y/N) ";AN$: IF LEFT$ (AN$,1) > < "Y" THEN GOTO 4520
4510 POKE 34,2: PRINT R$;D$;"PR#1": POKE 1401,80
4520 HOME :SP = PEEK (ACTSTL) * H + PEEK (ACTSTL - 1)
4530 FOR J = 0 TO NAD - 1
```

```
4540 PT = SP + J * A% * NC
4550 FOR I = 0 TO NC - 1
4560 B(I) = 0:PR = PT + I * A%
4570 FOR K = 0 TO A% - 1
4580 B(I) = B(I) + PEEK (PR + K) * (H ^ K) / DAS: NEXT K
4590 B(I) = B(I) * D(I) / 4095 / GA(I)
4600 NEXT I
4610 PRINT R$;J + 1;": ";
4620 FOR I = 0 TO NC - 1
4630 TP = I * 18 + 7:TP = TP - INT (TP / 40) * 40: IF PEEK (36) = > TP THEN TP = TP + 40
4640 PRINT TAB( TP);"C";I + 1;"="";B(I);
4650 NEXT I: NEXT J
4660 PRINT R$;D$;"PR#0": RETURN : END
6000 FOR I = 0 TO NC - 1
6010 G(I) = INT ( PEEK (GNCD + 4 - I) / 16):C(I) = PEEK (GNCD + 4 - I) - G(I) * 16:D(I) = G(I)
6020 IF D(I) > = 4 THEN D(I) = D(I) - 4
6030 B% = INT (D(I) / 2 + 0.5):C% = D(I) - INT (D(I) / 2) * 2
6040 D(I) = 5 * ((2 ^ C%) / (10 ^ B%))
6045 GA(I) = PEEK (G1R + I) / 100
6048 E(I) = PEEK (SVLTG + 2 * I) * H + PEEK (SVLTG + 2 * I + 1):E(I) = E(I) / 1E3
6050 NEXT I
6060 RETURN : END
9700 C = PEEK (116):D = PEEK (115):GOSUB 9900
9710 VTAB 3: PRINT "PRESENT HIMEM="; TAB( 33);A$
9720 VTAB 4: HTAB 32: PRINT "=";D + C * 256:Z1 = PEEK (37)
9730 VTAB 5: PRINT "WANT TO CHANGE HIMEM Y/N?": VTAB 5: HTAB 32: GET AN$
9740 IF AN$ > < "Y" THEN GOTO 9780
9750 VTAB 5: HTAB 1: PRINT "INPUT NEW HIMEM "
9755 VTAB 5: HTAB 17: INVERSE : PRINT "(4-HEX;$XXXX)=": NORMAL : VTAB 5: HTAB 32: INPUT A$
9760 GOSUB 9800:B = C * 256 + D: VTAB 6: HTAB 32: PRINT "=";B
9770 HIMEM: B:Z1 = PEEK (37)
9780 POKE 34,Z1: HOME : RETURN : END
9800 FOR I = 2 TO 5
9810 A(I) = ASC ( MID$ (A$,I,1))
9820 IF A(I) = < 57 THEN A(I) = A(I) - 48: GOTO 9840
9830 A(I) = A(I) - 55
9840 NEXT I
9850 C = A(2) * 16 + A(3):D = A(4) * 16 + A(5)
9860 RETURN : END
9900 A(1) = INT (C / 16):A(2) = C - A(1) * 16:A(3) = INT (D / 16):A(4) = D - A(3) * 16:A$ = "$"
9910 FOR I = 1 TO 4
9920 IF A(I) < 10 THEN A$ = A$ + STR$ (A(I)): GOTO 9940
9930 A(I) = A(I) + 55:A$ = A$ + CHR$ (A(I))
9940 NEXT I
9950 RETURN : END
```

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```
100 TEXT : HOME : FLASH : HTAB 7: PRINT "VOLTAGE--VELOCITY CONV": NORMAL
110 HIMEM: 36864: REM 36864=$9000
200 DIM A(9),B(9),C(19),D(9),E(9),G(9),GA(9)
205 DIM BV(255),AV(255),XR(255),YH(255),ZA(255),DAS(255),NAD(255),TAM(255),RG(255),VE(255),GT(255)
500 D$ = CHR$(4):R$ = CHR$(13)
1000 VTAB 3: INPUT "ENTER NATURAL CONV CONST = ";NT
1010 INPUT "ENTER FORCED CONV CONST = ";ZT
1020 GOSUB 4000: REM CAL-COEFF READING
1030 GOSUB 5000: REM OLD SYSASSY READING
1050 FOR I = 0 TO SZ - 1
1060 GOSUB 3000
1065 VE(I) = V2:GT(I) = GD
1070 GOSUB 2000: PRINT
1075 K = I
1080 INPUT "CONTINUE? (Y/N) ";AN$: IF LEFT$(AN$,1) > < "Y" THEN GOTO 1100
1090 NEXT I
1100 GOSUB 5500
1110 PRINT R$;D$;"RUN MENU,D1"
1120 END
2000 POKE 34,2: HOME : FOR J = 1 TO 39: PRINT "*";: NEXT J: PRINT
2010 PRINT "EXP NO= ";I + NO; TAB( 16);"COEFF FN= ";CF$: TAB( 41);"BV= ";BV(I);" V"; TAB( 21);"AV= ";AV(I);" V"
2020 PRINT "X/R= ";XR(I);" CM"; TAB( 21);"Y/O= ";YH(I);" CM"; TAB( 41);"Z= ";ZA(I);" CM"; TAB( 21);"REC GN= ";RG(I)
2030 PRINT "TOT DAT PTS= ";DAS(I);" * ";NAD(I);" = ";DAS(I) * NAD(I); TAB( 41);"TOT MEAS TM= ";TAM(I);" SEC"
2050 PRINT "MEAN VELOCITY= ";VE(I);" CM/SEC"; TAB( 41);"GRADIEN= ";GT(I);" (CM/SEC)/VOLT"
2060 FOR J = 1 TO 39: PRINT "*";: NEXT J: PRINT
2070 RETURN : END
3000 V1 = 1.39
3010 AL = .0016
3020 OH = .20
3030 DT = OH / AL
3040 RP = 5.01 * (1 + OH) - 0.2
3050 BT = RP / (10 + RP) ^ 2
3060 SN = (NT ^ .8) * (BT ^ (- .2)) * EO ^ (- .4)
3070 DR = DT / BT * (1 / BV(I) ^ 2 - 1 / EO ^ 2)
3080 SO = (NT ^ .8) * (BT ^ (- .2)) * BV(I) ^ (- .4)
3090 SC = SO - SN
3100 TC = SN / (ZT * V1 ^ (- .4) + SN) - SO / (ZT * V1 ^ (- .4) + SO)
3110 TS = TC * ZT * V1 ^ (- .4)
3120 DC = DR - SC - TS
3130 VG = 1 / AV(I) ^ 2 - BT * DC / DT
3140 VT = (1 / VG) ^ .5
3150 VN = VT - EO
3160 V2 = C(0) + C(1) * VN + C(2) * VN ^ 2 + C(3) * VN ^ 3 + C(4) * VN ^ 4
3170 ER = (V2 - V1) / V1:ER = ER * 100
3180 EL = 1
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```
3190 IF ABS (ER) > EL THEN V1 = V2
3200 IF ABS (ER) > EL THEN GOTO 3100
3210 GD = C(1) + 2 * C(2) * VN + 3 * C(3) * VN ^ 2 + 4 * C(4) * VN ^ 3
3220 RETURN : END
4000 INPUT "COEFF FILE NAME = ";CF$
4010 PRINT R$;D$;"OPEN ";CF$;"D2"
4020 PRINT R$;D$;"READ ";CF$
4030 INPUT C(0),C(1),C(2),C(3),C(4)
4040 INPUT EO
4050 PRINT R$;D$;"CLOSE ";CF$
4060 RETURN : END
5000 INPUT "SYS-ASSY FILE NAME = ";F$
5010 PRINT R$;D$;"OPEN ";F$;"D2"
5020 PRINT R$;D$;"READ ";F$
5030 INPUT SZ,NO,N9
5040 FOR I = 0 TO SZ - 1
5050 INPUT BV(I),AV(I),XR(I),YH(I),ZA(I),DAS(I),NAD(I),TAM(I),RG(I)
5060 NEXT I
5110 PRINT R$;D$;"CLOSE ";F$
5120 RETURN : END
5500 INPUT " REFINED SYS-ASSY FILE NAME= ";G$
5510 SZ = K + 1
5520 PRINT R$;D$;"OPEN ";G$;"D2"
5530 PRINT R$;D$;"DELETE ";G$
5540 PRINT R$;D$;"OPEN ";G$
5550 PRINT R$;D$;"WRITE ";G$
5560 PRINT SZ;"NO";"NO + SZ - 1";"CF$
5570 FOR I = 0 TO SZ - 1
5580 PRINT VE(I);"GT(I);"XR(I);"YH(I);"ZA(I);"DAS(I);"NAD(I);"TAM(I);"RG(I)
5590 NEXT I
5600 PRINT R$;D$;"CLOSE ";G$
5610 RETURN : END
6000 REM INPUT OF COORDINATES FROM THE AUTO-CORRELATION SPECTRUM
6010 HOME : HTAB 5: INVERSE : PRINT "AUTOCORRELATION ENTRY MODE": NORMAL
6020 VTAB 7
6030 PRINT "1"; TAB( 5);"TIME=";TC(1): PRINT TAB( 5);"AUTOCORR=";AC(1)
6040 PRINT
6050 PRINT "2"; TAB( 5);"TIME=";TC(2): PRINT TAB( 5);"AUTOCORR=";AC(2)
6060 PRINT
6070 PRINT "3"; TAB( 5);"RECORDER SCALE FACTOR=";SF
6080 PRINT : HTAB 5: PRINT "H FOR HELP
"
6090 NORMAL : PRINT
6100 PRINT TAB( 5);"X FOR NO CHANGE"
6110 VTAB 18
```

199

```
6120 HTAB 10: PRINT "WHICH? ": GET A$
6130 VTAB 20
6140 IF A$ = "1" THEN INPUT "1)..TIME,AUTOCORR= ";TC(1),AC(1)
6150 IF A$ = "1" THEN GOTO 6000
6160 IF A$ = "2" THEN INPUT "2)..TIME,AUTOCORR= ";TC(2),AC(2)
6170 IF A$ = "2" THEN GOTO 6000
6180 IF A$ = "3" THEN INPUT "SCALE FACTOR= ";SF
6190 IF A$ = "3" THEN GOTO 6000
6200 IF A$ = "X" THEN HOME
6210 IF A$ = "X" THEN RETURN
6220 IF A$ = "H" THEN GOSUB 7500
6230 IF A$ = "H" THEN GOTO 6000
6240 RETURN : END
6500 REM ...PRINT OUT SECTION FOR SPECTRAL CHARACTERISTICS
6510 PRINT : PRINT
6520 PRINT "MICROSCALE= ";LM(2);" CMS"
6530 PRINT : PRINT "FLUC.VELLOCITY, FROM AUTOCORR= ";TV(2);" CM/S"
6540 PRINT : PRINT "DISSIPATION ENERGY= ";EP(1);" WATTS"
6550 PRINT : PRINT "KINETIC ENERGY= ";KE(1);" WATTS"
6560 FOR I = 1 TO 35: PRINT "=";: NEXT I: PRINT
6570 RETURN : END
7000 REM ...CALCULATION OF SPECTRAL CHARACTERISTICS
7010 FT = AC(2) / AC(1)
7020 LM(1) = TC(2) / SQRT (2 * (1 - FT))
7030 LM(2) = LM(1) * VL(1)
7040 TV(2) = (AC(1) ^ .5) * GD
7050 TV(2) = TV(2) * SF
7060 NU = 2.87E - 7
7070 EP(1) = 15 * NU * TV(2) ^ 2 / (LM(2) ^ 2)
7080 KE(1) = (TV(2) * 1E - 2) ^ 2
7090 RETURN : END
7500 REM ..AUTOCORR READ OUT HELP SEGMENT
7510 HOME
7520 INPUT "MIN. ON TIME SCALE? ";XL(1)
7530 INPUT "MAX.ON TIME SCALE? ";XL(2)
7540 INPUT "SPAN IN CMS? ";XL(3)
7550 INPUT "DIST.YOU WISH TO INTERPRET? ";XL(4)
7560 INVERSE : PRINT "END OF TIME-AXIS INPUT": NORMAL
7570 INPUT "MIN.ON AUTOCORR SCALE? ";YL(1)
7580 INPUT "MAX.ON AUTOCORR SCALE? ";YL(2)
7590 INPUT "SPAN IN CMS? ";YL(3)
7600 INPUT "DIST.YOU WISH TO INTERPRET? ";YL(4)
7610 INVERSE : PRINT "END OF INPUT ": NORMAL
7620 XL(5) = (XL(2) - XL(1)) * XL(4) / XL(3)
7630 YL(5) = (YL(2) - YL(1)) * YL(4) / YL(3)
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200

7640 TC(1) = 0:TC(2) = XL(5)  
7650 AC(2) = YL(5) + YL(1)  
7660 RETURN : END



201

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100 REM THIS ROUTINE CONTROLS THE PLOTTER AND WRITES COORDINATE DATA TO DISK
110 TEXT : HOME : FLASH : HTAB 7: PRINT "SPECTRUM CO-OR SAVE": NORMAL
120 HIMEM: 36864: REM 36864=$9000
200 DIM SV%(1,1023),KL(255),ED(255),DD(255)
300 PTRED = 36992:CNTRL = 25
500 D$ = CHR$(4):R$ = CHR$(13):YS = 10.6
1000 PRINT R$;D$;"BLOAD PLTRED1.OBJO,D1"
1100 POKE 34,2: HOME : VTAB 3
1110 PRINT " 1.INSERT PLOT ON PLOTTER"
1120 PRINT " 2.LOCATE THE PEN AT ORIGIN"
1130 PRINT " 3.(R,D,F,C)=CTRL KEY FOR FAST PEN MOVE"
1140 PRINT " 4.(I,U,K,M)=CTRL KEY FOR SLOW PEN MOVE"
1150 PRINT " 5.(/)=STORE CO-OR INTO MEMORY"
1160 PRINT " 6.<CR>=END KEY FOR CO-OR READING"
1165 PRINT : HTAB 2: INVERSE : PRINT "1,10 HZ MK MUST BE READING #1 & #2": NORMAL
1170 PRINT : PRINT "PRESS ANY KEY TO START": GET AN$
1200 CALL PTRED
1300 TP = PEEK (CNTRL) + PEEK (CNTRL + 1) * 256
1310 NP = TP - 2
1320 PRINT TAB( 2);NP;" DATA POINTS ARE TAKEN"
1400 GOSUB 5000
1410 SZ = NP: GOSUB 3000: GOSUB 2000
1500 INPUT " WANT TO CONTINUE? (Y/N) ";AN$: IF LEFT$(AN$,1) > < "N" THEN GOTO 1100
1600 PRINT R$;D$;"RUN MENU,D1"
1700 END
2000 POKE 34,2: HOME : VTAB 3
2010 INPUT " WANT TO PRINT OUT? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 2030
2020 PRINT R$;D$;"PR#1": POKE 1401,80
2030 HOME : VTAB 3
2040 FOR I = 1 TO 39: PRINT "*";: NEXT I: PRINT
2050 PRINT "EXPERIMENT NO = ";XN
2060 PRINT "KINETIC ERG(VOLT^2) = ";KE
2070 PRINT "DISSIP ERG(VOLT^2) = ";EP
2080 FOR I = 1 TO 39: PRINT "*";: NEXT I: PRINT
2090 PRINT R$;D$;"PR#0"
2100 RETURN : END
3000 POKE 34,2: HOME : PRINT " INTEGRATING..."
3005 XO% = SV%(0,0):X1% = SV%(0,1)
3010 FOR I = 2 TO TP - 1
3020 KL(I - 1) = (SV%(0,I) - XO%) / (X1% - XO%)
3030 ED(I - 1) = EO + SV%(1,I) * (E2 - EO) / (YS * 78.74)
3040 DD(I - 1) = 20 * KL(I) + ED(I)
3050 NEXT I
3060 REM INEGRATE FOR KE AND EP
3070 REM INTEGRATION USES TRAPEZOIDAL RULE
```

202

```
3080 KE = 0.0
3090 EP = 0.0
3100 FOR I = 1 TO NP - 1
3110 DK = 10 ^ KL(I + 1) - 10 ^ KL(I)
3120 EM = (10 ^ (ED(I + 1) / 10) + 10 ^ (ED(I) / 10)) / 2
3130 DM = (10 ^ (DD(I + 1) / 10) + 10 ^ (DD(I) / 10)) / 2
3140 KE = KE + EM * DK
3150 EP = EP + DM * DK
3160 NEXT I
3170 RETURN : END
5000 REM SAVE COORDINATE SET TO DISK
5030 INPUT "ENTER MIN.,MAX ON DB SCALE ";EO,E2
5040 INPUT "EXPERIMENT # = ";F$
5050 F$ = "SPEC-" + F$
5060 YS = 5.5
5070 PRINT R$;D$;"OPEN ";F$;".D2"
5075 PRINT R$;D$;"DELETE ";F$
5080 PRINT R$;D$;"OPEN ";F$
5085 PRINT R$;D$;"WRITE ";F$
5090 PRINT NP
5100 PRINT EO;" ";E2
5110 FOR I = 0 TO TP - 1
5120 PRINT SV%(O,I);" ";SV%(1,I)
5130 NEXT I
5140 PRINT R$;D$;"CLOSE ";F$
5150 HOME : PRINT F$;".D2 SAVE COMPLETED"
5160 RETURN : END
```

203

```
100 TEXT : HOME : FLASH : HTAB 7: PRINT "SPECTRUM ANALYSIS": NORMAL
110 HIMEM: 36864: REM 36864=$9000
200 DIM SV%(1,255),KL(255),ED(255),DD(255),KE(255),EP(255)
500 D$ = CHR$(4):R$ = CHR$(13):YS = 10.6
1000 VTAB 3: INPUT "STARTING/FINAL EXP NO = ";NO,N9
1010 FOR XN = NO TO N9
1020 F$ = "SPEC-" + STR$(XN): GOSUB 5000: REM DATA READ
1030 GOSUB 3000: REM CALCULATION STEP
1040 GOSUB 2000: REM RESULT PRINT
1050 K = XN - NO:KE(K) = KE:EP(K) = EP
1060 INPUT " CONTINUE? (Y/N) ";AN$: IF LEFT$(AN$,1) > < "Y" THEN GOTO 1080
1070 NEXT XN
1080 GOSUB 5500
1090 PRINT R$;D$;"RUN MENU,D1"
1100 END
2000 POKE 34,2: HOME : VTAB 3
2010 INPUT " WANT TO PRINT OUT? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 2030
2020 PRINT R$;D$;"PR#1": POKE 1401,80
2030 HOME : VTAB 3
2040 FOR I = 1 TO 39: PRINT "*";: NEXT I: PRINT
2050 PRINT "EXPERIMENT NO = ";XN
2060 PRINT "KINETIC ERG(VOLT^2) = ";KE
2070 PRINT "DISSIP ERG(VOLT^2) = ";EP
2080 FOR I = 1 TO 39: PRINT "*";: NEXT I: PRINT
2090 PRINT R$;D$;"PR#0"
2100 RETURN : END
3000 POKE 34,2: HOME : VTAB 3: PRINT F$;" INTEGRATING..."
3010 FOR I = 1 TO SZ
3020 KL(I) = (SV%(0,I) - X0%) / (X1% - X0%)
3030 ED(I) = EO + SV%(1,I) * (E2 - EO) / (YS * 78.74)
3040 DD(I) = 20 * KL(I) + ED(I)
3050 NEXT I
3060 REM INEGRATE FOR KE AND EP
3070 REM INTEGRATION USES TRAPEZOIDAL RULE
3080 KE = 0.0
3090 EP = 0.0
3100 FOR I = 1 TO SZ - 1
3110 DK = 10 ^ KL(I + 1) - 10 ^ KL(I)
3120 EM = (10 ^ (ED(I + 1) / 10) + 10 ^ (ED(I) / 10)) / 2
3130 DM = (10 ^ (DD(I + 1) / 10) + 10 ^ (DD(I) / 10)) / 2
3140 KE = KE + EM * DK
3150 EP = EP + DM * DK
3160 NEXT I
3170 RETURN : END
5000 POKE 34,2: HOME : VTAB 3: PRINT F$;" READING..."
```

203

204

```
5010 PRINT R$;D$;"OPEN ";F$;".D2"
5020 PRINT R$;D$;"READ ";F$
5030 INPUT SZ
5040 INPUT EO,E2
5050 INPUT XO%,YO%
5060 INPUT X1%,Y1%
5070 FOR I = 1 TO SZ
5080 INPUT SV%(O,I),SV%(1,I)
5090 NEXT I
5100 PRINT R$;D$;"CLOSE ";F$
5110 RETURN : END
5500 INPUT "SPEC-ASSY FILE NAME = ";G$
5510 SZ = K + 1
5520 PRINT R$;D$;"OPEN ";G$;".D2"
5530 PRINT R$;D$;"DELETE ";G$
5540 PRINT R$;D$;"OPEN ";G$
5550 PRINT R$;D$;"WRITE ";G$
5560 PRINT SZ;",";NO;",";NO + SZ - 1
5570 FOR I = 0 TO SZ - 1
5580 PRINT KE(I);",";EP(I)
5590 NEXT I
5600 PRINT R$;D$;"CLOSE ";G$
5610 RETURN : END
```

205

```
100 TEXT : HOME : FLASH : HTAB 7: PRINT "K.E. & D.E. CALCULATION": NORMAL
110 HIMEM: 36864: REM 36864 = $9000
200 DIM A(9),B(9),C(19),D(9),E(9),G(9),GA(9)
205 DIM VE(255),GT(255),XR(255),YH(255),ZA(255),DAS(255),NAD(255),TAM(255),RG(255),KE(255),DE(255)
500 D$ = CHR$(4):R$ = CHR$(13)
1000 VTAB 3: GOSUB 4000: REM REF'NED SYS FILE READING
1010 GOSUB 5000: REM OLD SPECASY READING
1020 FOR I = 0 TO SZ - 1
1030 OK = KE(I):OD = DE(I)
1040 GOSUB 3000
1050 GOSUB 2000: PRINT
1055 K = I
1060 INPUT "CONTINUE? (Y/N) ";AN$: IF LEFT$(AN$,1) > < "Y" THEN GOTO 1080
1070 NEXT I
1080 GOSUB 5500
1090 PRINT R$;D$;"RUN MENU,D1"
1100 END
2000 POKE 34,2: HOME : FOR J = 1 TO 39: PRINT "*":; NEXT J: PRINT
2010 PRINT "EXP NO= ";I + NO; TAB( 16);"COEFF FN= ";CF$
2015 PRINT "KE= ";OK;" VOLT^2"; TAB( 41);"DE= ";OD;" VOLT^2"
2020 PRINT "X/R= ";XR(I);" CM"; TAB( 21);"Y/O= ";YH(I);" CM"; TAB( 41);"Z= ";ZA(I);" CM"; TAB( 21);"REC GN= ";RG(I)
2030 PRINT "TOT DAT PTS= ";DAS(I);" * ";NAD(I);" = ";DAS(I) * NAD(I); TAB( 41);"TOT MEAS TM= ";TAM(I);" SEC"
2040 PRINT "MEAN VELOCITY= ";VE(I);" CM/SEC"; TAB( 41);"GRADIENT= ";GT(I);" (CM/SEC)/VOLT"
2050 PRINT "KINETIC ERG= ";KE(I);" WATT"; TAB( 41);"DISSIP ERG= ";DE(I);" WATT"
2060 FOR J = 1 TO 39: PRINT "*":; NEXT J: PRINT
2070 RETURN : END
3000 KE(I) = (3 / 2) * ((GT(I) / 100 / RG(I)) ^ 2) * KE(I)
3100 DE(I) = (1.54E - 4 / ((VE(I) / 100) ^ 2)) * ((GT(I) / 100 / RG(I)) ^ 2) * DE(I)
3200 RETURN : END
4000 INPUT "REF'NED SYS FILE NAME= ";F$
4010 PRINT R$;D$;"OPEN ";F$;" ,D2"
4020 PRINT R$;D$;"READ ";F$
4030 INPUT SZ,NO,N9,CF$
4040 FOR I = 0 TO SZ - 1
4050 INPUT VE(I),GT(I),XR(I),YH(I),ZA(I),DAS(I),NAD(I),TAM(I),RG(I)
4060 NEXT I
4070 PRINT R$;D$;"CLOSE ";F$
4080 RETURN : END
5000 INPUT "SPEC-ASSY FILE NAME = ";H$
5010 PRINT R$;D$;"OPEN ";H$;" ,D2"
5020 PRINT R$;D$;"READ ";H$
5030 INPUT SZ,NO,N9
5040 FOR I = 0 TO SZ - 1
5050 INPUT KE(I),DE(I)
5060 NEXT I
```

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```
5110 PRINT R$;D$;"CLOSE ";F$
5120 RETURN : END
5500 INPUT " FINAL DATA FILE NAME= ";G$
5510 SZ = K + 1:SF = 1E3
5520 PRINT R$;D$;"OPEN ";G$;".D2"
5530 PRINT R$;D$;"DELETE ";G$
5540 PRINT R$;D$;"OPEN ";G$
5550 PRINT R$;D$;"WRITE ";G$
5560 Y$ = ""
5570 Z$ = STR$ (SZ):L = 6: GOSUB 8100
5580 Z$ = STR$ (NO):L = 6: GOSUB 8100
5590 Z$ = STR$ (NO + SZ - 1):L = 6: GOSUB 8100
5600 Z$ = STR$ (SF):L = 6: GOSUB 8100
5610 Z$ = CF$:L = 10: GOSUB 8100
5620 PRINT Y$
5630 FOR I = 0 TO SZ - 1
5640 Y$ = ""
5650 Z = XR(I):Z = Z * SF:L = 5: GOSUB 8000
5660 Z = YH(I):Z = Z * SF:L = 5: GOSUB 8000
5670 Z = ZA(I):Z = Z * SF:L = 5: GOSUB 8000
5680 Z = VE(I):L = 12: GOSUB 8500
5690 Z = KE(I):L = 12: GOSUB 8500
5700 Z = DE(I):L = 12: GOSUB 8500
5710 Z = DAS(I):L = 4: GOSUB 8000
5720 Z = NAD(I):L = 4: GOSUB 8000
5730 Z = TAM(I):L = 12: GOSUB 8500
5735 PRINT Y$
5740 NEXT I
5750 PRINT R$;D$;"CLOSE ";G$
5760 RETURN : END
8000 Z$ = STR$ (Z)
8100 IF LEN (Z$) = L THEN GOTO 8120
8110 FOR M = 1 TO L - LEN (Z$):Z$ = " " + Z$: NEXT M
8120 Y$ = Y$ + Z$
8130 RETURN : END
8500 IF Z = 0 THEN Z1$ = "0.":SG$ = "+":PO$ = "00":SP$ = "+": GOTO 8640
8510 IF SGN (Z) = - 1 THEN Z = ABS (Z):SG$ = "-": GOTO 8530
8520 SG$ = "+"
8530 X = LOG (Z) / LOG (10):X1 = ABS ( INT (X))
8540 ON SGN (X) + 2 GOTO 8550,8560,8570
8550 FOR XX = 1 TO X1:Z = Z * 10: NEXT XX:SP$ = "-": GOTO 8590
8560 SP$ = "+": GOTO 8590
8570 IF X1 = 0 THEN GOTO 8560
8580 FOR XX = 1 TO X1:Z = Z / 10: NEXT XX:SP$ = "+"
8590 IF LEN ( STR$ (Z)) = 1 THEN Z$ = STR$ (Z) + ".": GOTO 8610
```

207

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8600 Z$ = STR$ (Z)
8610 IF INT (X1 / 10) > 0 THEN PO$ = STR$ (X1): GOTO 8630
8620 PO$ = "0" + STR$ (X1)
8630 Z1$ = LEFT$ (Z$,L - 5)
8640 Z$ = SG$ + Z1$ + "E" + SP$ + PO$
8650 GOSUB 8100
8660 RETURN : END
```

208

```
100 TEXT : HOME : FLASH : HTAB 7: PRINT "MAKING NORMALIZED CALASSY": NORMAL
200 DIM BV(255),AV(255),VE(255),DR(255),CV(255),CE(255)
210 HIMEM: 36864: REM 36864=$9000
500 D$ = CHR$(4):R$ = CHR$(13)
1000 VTAB 3: INPUT "ENTER NATURAL CONV CONSTANT = ";NT
1010 INPUT "ENTER FORCED CONV CONSTANT = ";ZT
1020 INPUT "ENTER NEW BASE VOLTAGE = ";EO
1030 GOSUB 5000: REM OLD CALASSY READING
1040 GOSUB 3000: POKE 34,2: HOME
1050 FOR I = 0 TO SZ - 1
1060 GOSUB 3500
1070 GOSUB 2000: PRINT
1075 K = I
1080 INPUT "CONTINUE? (Y/N) ";ANS: IF LEFT$(ANS,1) = "N" THEN GOTO 1100
1090 NEXT I
1100 GOSUB 5500
1110 PRINT R$;D$;"RUN MENU,D1"
1120 END
2000 PRINT
2010 PRINT "XP= ";I + NO; TAB( 11);"BV= ";BV(I); TAB( 26);"AV= ";AV(I);
2020 PRINT TAB( 41);"EO= ";EO; TAB( 11);"CV= ";CV(I); TAB( 26);"VE= ";VE(I)
2030 RETURN : END
3000 AL = .0016
3010 OH = 0.20
3020 DT = OH / AL
3030 RP = 5.01 * (1 + OH) - .2
3040 BT = RP / (10 + RP) ^ 2
3050 SN = (NT ^ .8) * (BT ^ (-.2)) * EO ^ (-.4)
3060 RETURN : END
3500 DR(I) = DT / BT * (1 / BV(I) ^ 2 - 1 / EO ^ 2)
3510 SO = (NT ^ .8) * (BT ^ (-.2)) * BV(I) ^ (-.4)
3520 TC = SN / (ZT * VE(I) ^ (-.4) + SN) - SO / (ZT * VE(I) ^ (-.4) + SO)
3530 TC = TC * ZT * VE(I) ^ (-.4)
3540 SC = SO - SN
3550 DR(I) = DR(I) - SC - TC
3560 V2 = 1 / AV(I) ^ 2 - BT * DR(I) / DT
3570 CV(I) = (1 / V2) ^ .5
3590 RETURN : END
5000 INPUT "CAL-ASSY FILE NAME = ";F$
5010 PRINT R$;D$;"OPEN ";F$;".D2"
5020 PRINT R$;D$;"READ ";F$
5030 INPUT SZ,NO,N9
5040 FOR I = 0 TO SZ - 1
5050 INPUT BV(I),AV(I),VE(I)
5060 NEXT I
```



209

```
5110 PRINT R$;D$;"CLOSE ";F$
5120 RETURN : END
5500 INPUT "REFINED CAL DATA F.N.= ";G$
5505 SZ = K + 1
5510 FOR I = 0 TO SZ / 2 - 1
5530 AV(I) = (CV(2 * I) + CV(2 * I + 1)) / 2
5540 CE(I) = (VE(2 * I) + VE(2 * I + 1)) / 2
5550 NEXT I
5560 PRINT R$;D$;"OPEN ";G$;" ,D2"
5562 PRINT R$;D$;"DELETE ";G$
5564 PRINT R$;D$;"OPEN ";G$
5570 PRINT R$;D$;"WRITE ";G$
5580 PRINT EO;" ,";SZ / 2 + 1
5585 PRINT EO;" ,";"O."
5600 FOR I = 0 TO SZ / 2 - 1
5610 PRINT AV(I);" ,";CE(I)
5620 NEXT I
5630 PRINT R$;D$;"CLOSE ";G$
5640 RETURN : END
```

210

```
100 TEXT : HOME : FLASH : HTAB 7: PRINT "PLOTING CAL-CURVE": NORMAL
200 DIM PL(1,19),CR%(1,39),CP$(1,0),CM$(0)
210 DIM B%(9),A$(19),CV%(1,1023),A(1,15)
220 GOSUB 6700: REM HIMEM SETTING SUBR
230 START = 36992:FRAME = 37004:ZLNDRW = 37118:MKTIC = 37221
240 CAPTIC = 37397:CURVE = 37513:DOTPLT = 37617
250 COMMENT = 37662:ENPLT = 37711:LINDRW = 37826
260 RTT = 38338:WDT = 38340:HGT = 38342
270 TCNM = 38344:TSFT = 38346:TLNG = 38348:FLAG = 38350
280 XREF = 38353:YREF = 38355:XLIM = 38357:YLIM = 38359
290 XZERO = 38361:YZERO = 38363:CRARY = 38365:CPARY = 38367
300 CMARY = 38369:CVARY = 38371:XCRD = 38373:YCRD = 38375
310 CXH = 38377:CYH = 38379:LPNMH = 38381:SYMBOL = 38383
320 UPI = 200:H = 256:R$ = CHR$(13):D$ = CHR$(4)
1000 POKE 34,6: HOME : PRINT " PLT-DEFAULTS READ & PLOT START "
1010 PRINT R$;D$;"BLOAD BPLT1.OBJO,D1"
1020 GOSUB 3000: REM PLT-DEFAULTS READ
1030 REM INITIALIZE PLOTTER
1040 CALL START
1100 REM FRAME (PL(I,J :I=0,1;J=0,1))
1110 POKE 34,6: HOME : PRINT " FRAME DRAW ": VTAB 9
1120 I1 = 0:I2 = 1:J1 = 0:J2 = 1: GOSUB 4000: PRINT
1130 INPUT "WANT TO CHANGE ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1150
1140 GOSUB 4200: GOTO 1110
1150 GOSUB 7000: CALL FRAME: REM (XREF,YREF,XLIM,YLIM)
1200 REM ZLNDRW (PL(I,J :I=0,1;J=0,1,2,3,4))
1210 POKE 34,6: HOME : PRINT " ZERO AXIS DRAW ": VTAB 9
1220 I1 = 0:I2 = 1:J1 = 0:J2 = 4: GOSUB 4000: PRINT
1230 INPUT "WANT TO CHANGE ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1250
1240 GOSUB 4200: GOTO 1210
1250 GOSUB 7100: CALL ZLNDRW: REM ((X/Y)ZERO,(X/Y)REF,(X/Y)LIM,FLAG(X/Y))
1300 REM MKTIC (PL(I,J :I=0,1;J=0,1,2,3,4,5,6))
1310 POKE 34,6: HOME : PRINT " TIC MARK DRAW ": VTAB 9
1320 I1 = 0:I2 = 1:J1 = 0:J2 = 6: GOSUB 4000: PRINT
1330 INPUT "WANT TO CHANGE ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1350
1340 GOSUB 4200: GOTO 1310
1350 GOSUB 7200: CALL MKTIC: REM ((X/Y)ZERO,TCNM(X/Y),TSFT(X/Y),TLNG(X/Y),CRARY,CR%(J,K))
1400 REM CAPTIC (PL(I,J :I=0,1;J=0,2,3,4,7,8,9,10,11,12),CP$(J,0 :J=0,1))
1410 POKE 34,6: HOME : PRINT " TIC CAPT DRAW ": VTAB 9
1420 I1 = 0:I2 = 1:J1 = 0:J2 = 0: GOSUB 4000
1430 I1 = 0:I2 = 1:J1 = 2:J2 = 4: GOSUB 4000
1440 I1 = 0:I2 = 1:J1 = 7:J2 = 12: GOSUB 4000
1450 I1 = 0:I2 = 1: GOSUB 4100: PRINT
1460 INPUT "WANT TO CHANGE PL ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1480
1470 GOSUB 4200
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1480 INPUT "WANT TO CHANGE CP$ ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1500
1490 GOSUB 4300: GOTO 1410
1500 GOSUB 7400: CALL CAPTIC: REM (RTT(X,Y),WDT(X,Y),HGT(X,Y),(X/Y)ZERO,TCNM(X/Y),(CR/CP)ARY,CP$(I,O),CR%(J,K)
1600 REM REDATA (PL(I,J :I=0,1;J=13,14)
1610 POKE 34,6: HOME : PRINT " READ DATA ": VTAB 9
1620 I1 = 0:I2 = 1:J1 = 13:J2 = 14: GOSUB 4000: PRINT
1630 INPUT "WANT TO CHANGE ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1650
1640 GOSUB 4200: GOTO 1610
1650 GOSUB 3300
1700 REM CURVE/DOTPLT (PL(I,J :I=0;J=2,3,13,14,15,16),PL(1,17))
1710 POKE 34,6: HOME : PRINT " CURVE/DOT PLOT ": VTAB 9
1720 I1 = 0:I2 = 1:J1 = 2:J2 = 3: GOSUB 4000
1730 I1 = 0:I2 = 1:J1 = 13:J2 = 16: GOSUB 4000: PRINT TAB( 22);"PL(1,17) = ";PL(1,17): PRINT
1740 INPUT "WANT TO CHANGE ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1760
1750 GOSUB 4200: GOTO 1710
1760 A = 2: GOSUB 7700
1770 ON A GOTO 1780,1790
1780 CALL CURVE: GOTO 1795: REM (LPNMH,SYMBOL,RTT,WDT,HGT,CV%(I,K),CVARY)
1790 CALL DOTPLT: REM (LPNMH,SYMBOL,RTT,WDT,HGT,CV%(I,K),CVARY)
1795 INPUT "CONTINUE TO DOT-PLOT? (Y/N) ";AN$: IF LEFT$(AN$,1) = "Y" THEN GOTO 1600
1800 REM REDATA (PL(I,J :I=0,1;J=2,3,13,14)
1810 POKE 34,6: HOME : PRINT " READ DATA ": VTAB 9
1815 I1 = 0:I2 = 1:J1 = 2:J2 = 3: GOSUB 4000
1820 I1 = 0:I2 = 1:J1 = 13:J2 = 14: GOSUB 4000: PRINT
1830 INPUT "WANT TO CHANGE ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1850
1840 GOSUB 4200: GOTO 1810
1850 GOSUB 4500: REM CAL-COEFF FILE READING & CALC
1900 REM CURVE/DOTPLT (PL(I,J :I=0;J=2,3,13,14,15,16),PL(1,17))
1910 POKE 34,6: HOME : PRINT " CURVE/DOT PLOT ": VTAB 9
1920 I1 = 0:I2 = 1:J1 = 2:J2 = 3: GOSUB 4000
1930 I1 = 0:I2 = 1:J1 = 13:J2 = 16: GOSUB 4000: PRINT TAB( 22);"PL(1,17) = ";PL(1,17): PRINT
1940 INPUT "WANT TO CHANGE ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1960
1950 GOSUB 4200: GOTO 1910
1960 A = 1: FLASH : PRINT "PLOTTER CO-OR'S CALCULATING": NORMAL : GOSUB 7700
1970 ON A GOTO 1980,1990
1980 CALL CURVE: GOTO 1995: REM (LPNMH,SYMBOL,RTT,WDT,HGT,CV%(I,K),CVARY)
1990 CALL DOTPLT: REM (LPNMH,SYMBOL,RTT,WDT,HGT,CV%(I,K),CVARY)
1995 INPUT "CONTINUE TO CURVE-PLOT? (Y/N) ";AN$: IF LEFT$(AN$,1) = "Y" THEN GOTO 1800
2000 REM COMMENT (R,W,HT,X,Y,CM$(O))
2010 POKE 34,6: HOME : PRINT " COMMENT DRAW ": VTAB 9
2020 GOSUB 4400
2030 GOSUB 8000: CALL COMMENT: REM (RTT,WDT,HGT,(X/Y)CRD,CMARY,CM$(O))
2040 INPUT "WANT TO CONTINUE ? (Y/N) ";AN$: IF LEFT$(AN$,1) > < "N" THEN GOTO 2020
2200 POKE 34,6: HOME : PRINT " PLOT END & PLT-DEFAULTS SAVE "
2210 CALL ENPLT
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2220 GOSUB 3100: REM PLT-DEFAULTS SAVE
2230 PRINT R$;D$;"RUN MENU,D1"
2240 END
2500 REM GENERAL LINE DRAW
2510 POKE 34,6: HOME : PRINT " GENERAL LINE DRAW": POKE 34,8: VTAB 9
2520 INPUT " A(0,11),A(1,11),A(0,12),A(1,12):RAW VALUE = ";A(0,11),A(1,11),A(0,12),A(1,12)
2530 GOSUB 8100: CALL LINDRW
2540 INPUT "CONTINUE ? (Y/N) ";AN$
2550 IF LEFT$(AN$,1) > < "N" GOTO 2520
3000 FM$ = "CALDEF"
3010 PRINT R$;D$;"OPEN ";FM$;".D1": PRINT R$;D$;"READ ";FM$
3020 FOR J = 0 TO 19
3030 INPUT PL(O,J),PL(1,J)
3040 NEXT J
3050 INPUT CP$(O,O): INPUT CP$(1,O)
3060 PRINT R$;D$;"CLOSE ";FM$
3070 RETURN : END
3100 FM$ = "CALDEF"
3110 PRINT R$;D$;"OPEN ";FM$;".D1": PRINT R$;D$;"DELETE ";FM$
3120 PRINT R$;D$;"OPEN ";FM$;".D1": PRINT R$;D$;"WRITE ";FM$
3130 FOR J = 0 TO 19: PRINT PL(O,J);", ";PL(1,J): NEXT J
3140 PRINT CP$(O,O): PRINT CP$(1,O)
3150 PRINT R$;D$;"CLOSE ";FM$
3160 RETURN : END
3300 REM REDATA (PL(I,J :I=0,1;J=13,14)
3310 POKE 34, PEEK (37) - 1: HOME
3320 INPUT "REF'NED CAL FILE NAME = ";F$
3330 PRINT R$;D$;"OPEN ";F$;".D2"
3340 PRINT R$;D$;"READ ";F$
3350 INPUT EO,SZ
3360 FOR K = 0 TO SZ - 1
3370 INPUT A(O,1),A(1,1)
3380 A(O,1) = A(O,1) - EO
3390 CV%(O,K) = INT (A(O,1) * PL(O,14) * PL(O,13) + 0.5)
3400 CV%(1,K) = INT (A(1,1) * PL(1,14) * PL(1,13) + 0.5)
3410 NEXT K
3420 PRINT R$;D$;"CLOSE ";F$
3430 PL(O,15) = SZ
3440 RETURN : END
4000 FOR J = J1 TO J2
4010 FOR I = I1 TO I2
4020 PRINT TAB( 2 + I * 20);"PL(";I;",";J;) = ";PL(I,J);
4030 NEXT I: PRINT
4040 NEXT J
4050 RETURN : END
```

```
4100 FOR I = I1 TO I2: PRINT "CP$( ";I;" ,O) = ";CP$(I,O): NEXT I: RETURN : END
4200 POKE 34, PEEK (37) - 1: HOME
4210 INPUT "J ARG OF (I,J) OR CTRL-S = ";J$
4220 IF J$ = CHR$(19) THEN RETURN : END
4230 INPUT "I ARG OF (I,J) & VALUE:I,V = ";I$,V$
4240 I = VAL (I$):J = VAL (J$):V = VAL (V$)
4250 PL(I,J) = V: GOTO 4210
4260 RETURN : END
4300 POKE 34, PEEK (37) - 1: HOME
4310 INPUT "I ARG OF (I,J) OR CTRL-S = ";I$
4320 IF I$ = CHR$(19) THEN RETURN : END
4330 INPUT "THEN TYPE STRING ";V$
4340 I = VAL (I$):CP$(I,O) = V$: GOTO 4310
4350 RETURN : END
4400 REM
4410 INPUT "ROT,WDT,HGT = ";R,W,HT
4420 INPUT "X,Y INCH = ";X,Y
4430 INPUT "CM$(O) = ";CM$(O)
4440 RETURN : END
4500 REM REDATA (PL(I,J :I=0,1;J=2,3,13,14)
4510 POKE 34, PEEK (37) - 1: HOME
4520 INPUT "CAL-COEFF FILE NAME = ";F$
4530 PRINT R$;D$;"OPEN ";F$:".D2"
4540 PRINT R$;D$;"READ ";F$
4550 INPUT A(O,10),A(O,11),A(O,12),A(O,13),A(O,14)
4560 INPUT EO
4570 PRINT R$;D$;"CLOSE ";F$
4580 INPUT "# OF POLY-FITTED POINTS= ";SZ
4590 IC = (PL(O,3) - PL(O,2)) / PL(O,14) / SZ
4600 FLASH : PRINT "READING VALUES...": NORMAL
4610 FOR K = 0 TO SZ
4620 A(O,1) = PL(O,3) / PL(O,13) + K * IC
4630 A(1,1) = A(O,10) + A(O,11) * A(O,1) + A(O,12) * A(O,1) ^ 2 + A(O,13) * A(O,1) ^ 3 + A(O,14) * A(O,1) ^ 4
4640 CV%(O,K) = INT (A(O,1) * PL(O,14) * PL(O,13) + 0.5)
4650 CV%(1,K) = INT (A(1,1) * PL(1,14) * PL(1,13) + 0.5)
4660 NEXT K
4670 PL(O,15) = SZ + 1
4680 RETURN : END
6700 C = PEEK (116):D = PEEK (115):GOSUB 6900
6710 VTAB 3: PRINT "PRESENT HIMEM="; TAB( 33);A$
6720 VTAB 4: HTAB 32: PRINT "=";D + C * 256:Z1 = PEEK (37)
6730 VTAB 5: PRINT "WANT TO CHANGE HIMEM Y/N = ": VTAB 5: HTAB 28: INPUT AN$
6740 IF AN$ > < "Y" THEN GOTO 6780
6750 VTAB 5: HTAB 1: PRINT "INPUT NEW HIMEM "
6755 VTAB 5: HTAB 17: INVERSE : PRINT "(4-HEX;$XXXX)=": NORMAL : VTAB 5: HTAB 32: INPUT A$
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6760 GOSUB 6800:B = C * 256 + D: VTAB 6: HTAB 32: PRINT "=";B
6770 HIMEM: B:Z1 = PEEK (37)
6780 POKE 34,Z1: HOME : RETURN : END
6800 FOR I = 2 TO 5
6810 A(O,I) = ASC ( MID$ (A$,I,1))
6820 IF A(O,I) = < 57 THEN A(O,I) = A(O,I) - 48: GOTO 6840
6830 A(O,I) = A(O,I) - 55
6840 NEXT I
6850 C = A(O,2) * 16 + A(O,3):D = A(O,4) * 16 + A(O,5)
6860 RETURN : END
6900 A(O,1) = INT (C / 16):A(O,2) = C - A(O,1) * 16:A(O,3) = INT (D / 16):A(O,4) = D - A(O,3) * 16:A$ = "$"
6910 FOR I = 1 TO 4
6920 IF A(O,I) < 10 THEN A$ = A$ + STR$ (A(O,I)): GOTO 6940
6930 A(O,I) = A(O,I) + 55:A$ = A$ + CHR$ (A(O,I))
6940 NEXT I
6950 RETURN : END
7000 REM SUBR FRAME (PL(I,J :I=0,1;J=0,1)
7010 B%(O) = INT (PL(O,0) * UPI) - 1: REM XREF
7020 B%(1) = INT (PL(1,0) * UPI) - 1: REM YREF
7030 B%(2) = INT ((PL(O,0) + PL(O,1)) * UPI) - 1: REM XLIM
7040 B%(3) = INT ((PL(1,0) + PL(1,1)) * UPI) - 1: REM YLIM
7050 FOR I = 0 TO 3
7060 POKE XREF + I * 2, INT (B%(I) / H): POKE XREF + I * 2 - 1,B%(I) - INT (B%(I) / H) * H
7070 NEXT I
7080 RETURN : END
7100 REM SUBR ZLNRW (PL(I,J :I=0,1;J=0,1,2,3,4)
7110 FOR I = 0 TO 1
7120 IF PL(I,2) * PL(I,3) = > 0 THEN POKE FLAG + I,0:B%(I) = INT (PL(I,0) * UPI) - 1: GOTO 7140
7130 POKE FLAG + I,1:B%(I) = INT ((PL(I,0) - PL(I,2) * PL(I,1) / (PL(I,3) - PL(I,2))) * UPI) - 1
7140 POKE XZERO + I * 2, INT (B%(I) / H): POKE XZERO + I * 2 - 1,B%(I) - INT (B%(I) / H) * H
7150 NEXT I
7160 RETURN : END
7200 REM SUBR MKTIC (PL(I,J :I=0,1;J=0,1,2,3,4,5,6)
7202 FOR I = 0 TO 1
7204 J = 1 - I
7206 POKE TSFT + J, INT (PL(J,5) * UPI): POKE TLNG + J,H - INT (PL(J,6) * UPI)
7208 NEXT I
7210 FOR I = 0 TO 1
7220 IF PL(I,2) * PL(I,3) = > 0 THEN B%(I) = INT (PL(I,0) * UPI) - 1: GOTO 7240
7230 B%(I) = INT ((PL(I,0) - PL(I,2) * PL(I,1) / (PL(I,3) - PL(I,2))) * UPI) - 1
7240 POKE XZERO + I * 2, INT (B%(I) / H): POKE XZERO + I * 2 - 1,B%(I) - INT (B%(I) / H) * H
7250 J = 1 - I
7260 B%(I + 2) = INT ((PL(J,3) - PL(J,2)) / PL(J,4)): POKE TCNM + J,B%(I + 2)
7270 FOR K = 0 TO B%(I + 2)
7280 CR%(J,K) = INT ((PL(J,0) + PL(J,1) * PL(J,4) * K / ABS (PL(J,3) - PL(J,2))) * UPI) - 1

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7290 NEXT K
7300 NEXT I
7310 B%(5) = PEEK (107) + PEEK (108) * H: REM PL(1,19) ARRAY ADDR
7320 B%(5) = B%(5) + 209 + 9 - 1: REM ADDR OF 1ST ELEMENT OF CR% - 1
7330 POKE CRARY, INT (B%(5) / H): POKE CRARY - 1, B%(5) - INT (B%(5) / H) * H
7340 RETURN : END
7400 REM SUBR CAPTIC (PL(I, J : I=0, 1; J=0, 2, 3, 4, 7, 8, 9, 10, 11, 12), CP$(J, 0 : J=0, 1))
7410 FOR I = 0 TO 1
7420 J = 1 - I
7430 B%(I + 2) = INT ((PL(J, 3) - PL(J, 2)) / PL(J, 4) / PL(J, 12))
7440 POKE TCNM + J, B%(I + 2): POKE RTT + J, PL(J, 7)
7450 POKE WDT + J, PL(J, 8)
7460 POKE HGT + J, PL(J, 9)
7470 NEXT I
7480 FOR I = 0 TO 1
7490 J = 1 - I
7500 IF PL(I, 2) * PL(I, 3) = > 0 THEN B%(I) = INT ((PL(I, 0) - PL(J, 10)) * UPI) - 1: GOTO 7520
7510 B%(I) = INT ((PL(I, 0) - PL(I, 2) * PL(I, 1) / (PL(I, 3) - PL(I, 2)) - PL(J, 10)) * UPI) - 1
7520 POKE XZERO + I * 2, INT (B%(I) / H): POKE XZERO + I * 2 - 1, B%(I) - INT (B%(I) / H) * H
7530 FOR K = 0 TO B%(I + 2)
7540 CR%(J, K) = INT ((PL(J, 0) + PL(J, 1) * PL(J, 4) * PL(J, 12) * K / ABS (PL(J, 3) - PL(J, 2)) - PL(J, 11)) * UPI) - 1
7550 NEXT K
7560 NEXT I
7570 B%(5) = PEEK (107) + PEEK (108) * H: REM PL(1,19) ARRAY ADDR
7580 B%(5) = B%(5) + 209 + 9 - 1: REM ADDR OF 1ST ELEMENT OF CR% - 1
7590 POKE CRARY, INT (B%(5) / H): POKE CRARY - 1, B%(5) - INT (B%(5) / H) * H
7600 B%(6) = B%(5) + 169: REM ADDR OF 1ST ELEMENT OF CP$ - 1
7610 POKE CPARY, INT (B%(6) / H): POKE CPARY - 1, B%(6) - INT (B%(6) / H) * H
7620 RETURN : END
7700 REM SUBR CURVE/DOTPLT (PL(I, J : I=0, 1; J=2, 3, 13, 14, 15, 16), PL(1, 17))
7702 POKE SYMBOL, PL(0, 16): POKE RTT, PL(1, 15)
7704 POKE WDT, PL(1, 16): POKE HGT, PL(1, 16)
7710 FOR K = 0 TO PL(0, 15) - 1
7720 IF (CV%(0, K) / PL(0, 13)) = > PL(0, 2) THEN K1 = K: GOTO 7740
7730 NEXT K
7740 FOR K = PL(0, 15) - 1 TO K1 STEP - 1
7750 IF (CV%(0, K) / PL(0, 13)) = < PL(0, 3) THEN K2 = K: GOTO 7770
7760 NEXT K
7770 B%(5) = K2 - K1 + 1
7780 POKE LPNMH, INT (B%(5) / H): POKE LPNMH - 1, B%(5) - INT (B%(5) / H) * H
7790 B%(0) = 2000: B%(1) = 1400: B%(2) = 1999: B%(3) = 1399
7800 FOR K = K1 TO K2
7810 FOR I = 0 TO 1
7820 CV%(I, K) = INT ((PL(I, 0) + (CV%(I, K) / PL(I, 13) - PL(I, 2)) * PL(I, 1) / (PL(I, 3) - PL(I, 2))) * UPI) - 1
7830 IF CV%(I, K) = > B%(I) THEN CV%(I, K) = B%(I + 2)

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7840 NEXT I
7850 NEXT K
7860 B%(5) = PEEK (107) + PEEK (108) * H: REM      PL(1,19) ARRAY ADDR
7870 B%(5) = B%(5) + 209 + 169 + 15 + 10 + 27 + 67 + 9: REM      ADDR OF 1ST ELEMENT OF CV%
7880 B%(6) = B%(5) + 4 * K1 - 1: REM      ADDR OF K1TH ELEMENT OF CV% -1
7890 POKE CVARY, INT (B%(6) / H): POKE CVARY - 1,B%(6) - INT (B%(6) / H) * H
7900 RETURN : END
8000 REM      COMMENT (R,W,HT,X,Y,CM$(O))
8010 POKE RTT,R: POKE WDT,W: POKE HGT,HT
8020 B%(O) = INT (X * UPI) - 1:B%(1) = INT (Y * UPI) - 1
8030 FOR I = 0 TO 1
8040 POKE XCRD + 2 * I, INT (B%(I) / H): POKE XCRD + 2 * I - 1,B%(I) - INT (B%(I) / H) * H
8050 NEXT I
8060 B%(5) = PEEK (107) + PEEK (108) * H: REM      PL(1,19) ARRAY ADDR
8070 B%(5) = B%(5) + 209 + 169 + 15 + 7 - 1: REM      ADDR OF 1ST ELEMENT OF CM$-1
8080 POKE CMARY, INT (B%(5) / H): POKE CMARY - 1,B%(5) - INT (B%(5) / H) * H
8090 RETURN : END
8100 REM      LINDRW (A(I,J:I=0,1;J=11,12),PL(I,J:I=0,1;J=0,1,2,3,14))
8110 FOR K = 11 TO 12
8120 FOR I = 0 TO 1
8130 J = (K - 11) * 2 + I
8140 B%(J) = INT ((PL(I,0) + (A(I,K) * PL(I,14) - PL(I,2)) * PL(I,1) / (PL(I,3) - PL(I,2))) * UPI) - 1
8150 NEXT I
8160 NEXT K
8170 FOR I = 0 TO 3: POKE XCRD + I * 2, INT (B%(I) / H): POKE XCRD + I * 2 - 1,B%(I) - INT (B%(I) / H) * H: NEXT I
8180 RETURN : END
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100 TEXT : HOME : FLASH : HTAB 7: PRINT "PLOTING PROBE-LOC": NORMAL
200 DIM PL(1,19),CR%(1,39),CP$(1,0),CM$(0)
210 DIM B%(9),A$(19),CV%(1,1023),A(1,15)
220 GOSUB 6700: REM HIMEM SETTING SUBR
230 START = 36992:FRAME = 37004:ZLNDRW = 37118:MKTIC = 37221
240 CAPTIC = 37397:CURVE = 37513:DOTPLT = 37617
250 COMMENT = 37662:ENPLT = 37711:LINDRW = 37826
260 RTT = 38338:WDT = 38340:HGT = 38342
270 TCNM = 38344:TSFT = 38346:TLNG = 38348:FLAG = 38350
280 XREF = 38353:YREF = 38355:XLIM = 38357:YLIM = 38359
290 XZERO = 38361:YZERO = 38363:CRARY = 38365:CPARY = 38367
300 CMARY = 38369:CVARY = 38371:XCRD = 38373:YCRD = 38375
310 CXH = 38377:CYH = 38379:LPNMH = 38381:SYMBOL = 38383
320 UPI = 200:H = 256:R$ = CHR$(13):D$ = CHR$(4)
500 GOSUB 4500: REM BATH-DEFAULTS READ
510 POKE 34,6: HOME : VTAB 7: PRINT " BATH DEFUALTS": VTAB 9
530 I1 = 0:I2 = 1:J1 = 1:J2 = 6: GOSUB 4700: PRINT
540 INPUT " WANT TO CHANGE ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 560
550 GOSUB 4800: GOTO 510
560 GOSUB 5000: REM GRID PT-->VLTG CONV
600 POKE 34,6: HOME : PRINT " PLT-DEFAULTS READ & PLOT START "
610 PRINT R$:D$:"BLOAD BPLT1.OBJO,D1"
620 GOSUB 3000: REM PLT-DEFAULTS READ
630 GOSUB 5300: REM A(I,J)-->PL(I,J)
1030 REM INITIALIZE PLOTTER
1040 CALL START
1100 REM FRAME (PL(I,J :I=0,1;J=0,1))
1110 POKE 34,6: HOME : PRINT " FRAME DRAW ": VTAB 9
1120 I1 = 0:I2 = 1:J1 = 0:J2 = 1: GOSUB 4000: PRINT
1130 INPUT "WANT TO CHANGE ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1150
1140 GOSUB 4200: GOTO 1110
1150 GOSUB 7000: CALL FRAME: REM (XREF,YREF,XLIM,YLIM)
1300 REM MKTIC (PL(I,J :I=0,1;J=0,1,2,3,4,5,6)
1310 POKE 34,6: HOME : PRINT " TIC MARK DRAW ": VTAB 9
1320 I1 = 0:I2 = 1:J1 = 0:J2 = 6: GOSUB 4000: PRINT
1330 INPUT "WANT TO CHANGE ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1350
1340 GOSUB 4200: GOTO 1310
1350 GOSUB 7200: CALL MKTIC: REM ((X/Y)ZERO,TCNM(X/Y),TSFT(X/Y),TLNG(X/Y),CRARY,CR%(J,K))
1400 REM CAPTIC (PL(I,J :I=0,1;J=0,2,3,4,7,8,9,10,11,12),CP$(J,0 :J=0,1))
1410 POKE 34,6: HOME : PRINT " TIC CAPT DRAW ": VTAB 9
1420 I1 = 0:I2 = 1:J1 = 0:J2 = 0: GOSUB 4000
1430 I1 = 0:I2 = 1:J1 = 2:J2 = 4: GOSUB 4000
1440 I1 = 0:I2 = 1:J1 = 7:J2 = 12: GOSUB 4000
1450 I1 = 0:I2 = 1: GOSUB 4100: PRINT
1460 INPUT "WANT TO CHANGE PL ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1480
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1470 GOSUB 4200
1480 INPUT "WANT TO CHANGE CP$ ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1500
1490 GOSUB 4300: GOTO 1410
1500 GOSUB 7400: CALL CAPTIC: REM (RTT(X,Y),WDT(X,Y),HGT(X,Y),(X/Y)ZERO,TCNM(X/Y),(CR/CP)ARY,CP$(I,O),CR%(J,K)
1600 REM ELECTRODE SURFACE LINE DRAW
1610 POKE 34,6: HOME : PRINT "ELECTRODE DRAW"
1620 A(O,11) = O:A(1,11) = A(1,1) - A(1,2):A(O,12) = A(O,2) / 2:A(1,12) = A(1,11)
1630 GOSUB 8100: CALL LINDRW
1640 A(O,11) = A(O,12):A(1,11) = A(1,12):A(O,12) = A(O,12):A(1,12) = A(1,1)
1650 GOSUB 8100: CALL LINDRW
1700 REM COMMENT (R,W,HT,X,Y,CM$(O))
1710 POKE 34,6: HOME : PRINT " COMMENT DRAW ": VTAB 9
1720 GOSUB 4400
1730 GOSUB 8000: CALL COMMENT: REM (RTT,WDT,HGT,(X/Y)CRD,CMARY,CM$(O))
1740 INPUT "WANT TO CONTINUE ? (Y/N) ";AN$: IF LEFT$(AN$,1) > < "N" THEN GOTO 1720
1800 REM REDATA (PL(I,J :I=O,1;J=13,14)
1810 POKE 34,6: HOME : PRINT " READ DATA ": VTAB 9
1815 INPUT " INITIAL POINT # = ";PNR:PNR = PNR - 1
1820 I1 = O:I2 = 1:J1 = 13:J2 = 14: GOSUB 4000: PRINT
1830 INPUT "WANT TO CHANGE ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1850
1840 GOSUB 4200: GOTO 1810
1850 GOSUB 3300
1900 REM CURVE/DOTPLT (PL(I,J :I=O;J=2,3,13,14,15,16),PL(1,17))
1910 POKE 34,6: HOME : PRINT " CURVE/DOT PLOT ": VTAB 9
1920 I1 = O:I2 = 1:J1 = 2:J2 = 3: GOSUB 4000
1930 I1 = O:I2 = 1:J1 = 13:J2 = 16: GOSUB 4000
1935 PRINT TAB( 2);"PT NO=" ;PNR + 1; TAB( 22);"PL(1,17)=" ;PL(1,17): PRINT
1940 INPUT "WANT TO CHANGE ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 1960
1950 GOSUB 4200: GOTO 1910
1960 A = 2: GOSUB 7700
1970 ON A GOTO 1980,1990
1980 CALL CURVE: GOTO 1700: REM (LPNMH,SYMBOL,RTT,WDT,HGT,CV%(I,K),CVARY)
1990 CALL DOTPLT: REM (LPNMH,SYMBOL,RTT,WDT,HGT,CV%(I,K),CVARY)
2000 B%(O) = CV%(O,O) - 40:B%(1) = CV%(1,O) - 40:PNR = PNR + 1
2010 POKE RTT,8: POKE WDT,2: POKE HGT,2
2020 CM$(O) = STR$(PNR)
2030 GOSUB 8030: CALL COMMENT
2090 POKE 34, PEEK (37) - 1: HOME
2100 INPUT " WANT TO CONTINUE? (Y/N) ";AN$: IF LEFT$(AN$,1) > < "N" THEN GOTO 1850
2110 INPUT " CONTINUE W/OTHER INDENT? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 2200
2120 POKE 34,6: HOME : VTAB 7: PRINT " BATH DEFUALTS": VTAB 9
2130 I1 = O:I2 = 1:J1 = 1:J2 = 6: GOSUB 4700: PRINT
2140 INPUT " WANT TO CHANGE ? (Y/N) ";AN$: IF LEFT$(AN$,1) = "N" THEN GOTO 2160
2150 GOSUB 4800: GOTO 2120
2160 GOSUB 5000: REM GRID PT-->VLTG CONV
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2170 GOTO 1800
2200 POKE 34,6: HOME : PRINT " PLOT END & PLT-DEFAULTS SAVE "
2210 CALL ENPLT
2220 GOSUB 3100: REM PLT-DEFAULTS SAVE
2230 GOSUB 4600: REM BATH-DEFAULTS SAVE
2240 PRINT R$;D$;"RUN MENU,D1"
2250 END
3000 PRINT R$;D$;"OPEN BHPLDEF,D1"
3010 PRINT R$;D$;"READ BHPLDEF"
3020 FOR J = 0 TO 19
3030 INPUT PL(O,J),PL(1,J)
3040 NEXT J
3050 INPUT CP$(O,O): INPUT CP$(1,O)
3060 PRINT R$;D$;"CLOSE BHPLDEF"
3070 RETURN : END
3100 PRINT R$;D$;"OPEN BHPLDEF,D1"
3110 PRINT R$;D$;"DELETE BHPLDEF"
3120 PRINT R$;D$;"OPEN BHPLDEF"
3130 PRINT R$;D$;"WRITE BHPLDEF"
3140 FOR J = 0 TO 19: PRINT PL(O,J);", ";PL(1,J): NEXT J
3150 PRINT CP$(O,O): PRINT CP$(1,O)
3160 PRINT R$;D$;"CLOSE BHPLDEF"
3170 RETURN : END
3300 REM REDATA (PL(I,J :I=0,1;J=13,14)
3310 POKE 34, PEEK (37) - 1: HOME
3320 INPUT " ENTER (1.VLTG 2.LOC): WHICH? (1/2) ";CH: ON CH GOTO 3330,3340
3330 INPUT " TYPE RADIAL/AXIAL VLTG ";RVT,ZVT: GOSUB 5200: GOTO 3350
3340 INPUT " TYPE RADIAL/AXIAL LOC ";RL,ZL: GOSUB 5400
3350 HOME :: GOSUB 5500
3380 CV%(O,O) = INT (RL * PL(O,14) * PL(O,13) + 0.5)
3390 CV%(1,O) = INT (ZL * PL(1,14) * PL(1,13) + 0.5)
3420 PL(O,15) = 1
3430 RETURN : END
4000 FOR J = J1 TO J2
4010 FOR I = I1 TO I2
4020 PRINT TAB( 2 + I * 20);"PL(";I;",";J;) = ";PL(I,J);
4030 NEXT I: PRINT
4040 NEXT J
4050 RETURN : END
4100 FOR I = I1 TO I2: PRINT "CP$(";I;",";O) = ";CP$(I,O): NEXT I: RETURN : END
4200 POKE 34, PEEK (37) - 1: HOME
4210 INPUT "J ARG OF (I,J) OR CTRL-S = ";J$
4220 IF J$ = CHR$ (19) THEN RETURN : END
4230 INPUT "I ARG OF (I,J) & VALUE:I,V = ";I$,V$
4240 I = VAL (I$):J = VAL (J$):V = VAL (V$)
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4250 PL(I,J) = V: GOTO 4210
4260 RETURN : END
4300 POKE 34, PEEK (37) - 1: HOME
4310 INPUT "I ARG OF (I,J) OR CTRL-S = ";I$
4320 IF I$ = CHR$ (19) THEN RETURN : END
4330 INPUT "THEN TYPE STRING ";V$
4340 I = VAL (I$):CP$(I,O) = V$: GOTO 4310
4350 RETURN : END
4400 REM
4410 INPUT "ROT,WDT,HGT = ";R,W,HT
4420 INPUT "X,Y INCH = ";X,Y
4430 INPUT "CM$(O) = ";CM$(O)
4440 RETURN : END
4500 PRINT R$;D$;"OPEN BATHDEF,D1"
4510 PRINT R$;D$;"READ BATHDEF"
4530 FOR J = 1 TO 6: INPUT A(O,J),A(1,J): NEXT J
4540 PRINT R$;D$;"CLOSE BATHDEF"
4550 RETURN : END
4600 PRINT R$;D$;"OPEN BATHDEF,D1"
4610 PRINT R$;D$;"DELETE BATHDEF"
4620 PRINT R$;D$;"OPEN BATHDEF"
4630 PRINT R$;D$;"WRITE BATHDEF"
4650 FOR J = 1 TO 6: PRINT A(O,J);",":A(1,J): NEXT J
4660 PRINT R$;D$;"CLOSE BATHDEF"
4670 RETURN : END
4700 FOR J = J1 TO J2
4710 FOR I = I1 TO I2
4720 PRINT TAB( 2 + I * 20);"A(";I;",";J;) = ";A(I,J);
4730 NEXT I: PRINT
4740 NEXT J
4750 RETURN : END
4800 POKE 34, PEEK (37) - 1: HOME
4810 INPUT "J ARG OF (I,J) OR CTRL-S = ";J$
4820 IF J$ = CHR$ (19) THEN RETURN : END
4830 INPUT "I ARG OF (I,J) & VALUE:I,V = ";I$,V$
4840 I = VAL (I$):J = VAL (J$):V = VAL (V$)
4850 A(I,J) = V: GOTO 4810
4860 RETURN : END
5000 POKE 34,6: HOME : PRINT " GRID PT-->VLTG CONV": PRINT : VTAB 9
5010 INPUT " WANT TO PRINT OUT? (Y/N) ":AN$: IF LEFT$ (AN$,1) > < "Y" THEN GOTO 5025
5020 PRINT R$;D$;"PR#1": POKE 1401,80:
5025 POKE 34,8: HOME
5030 INPUT " # OF MESHES (RAD/AX)IAL = ";NR,NZ
5040 DR = (A(O,6) - A(O,5)) / NR:DZ = (A(1,6) - A(1,5)) / NZ
5048 PRINT "RI= ";A(O,3); TAB( 21);"AI= ";A(1,3); TAB( 41);"RVTO= ";A(O,4); TAB( 21);"ZVTO= ";A(1,4)
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5050 FOR J = 0 TO NZ
5055 FOR K = 0 TO 39: PRINT "*";: NEXT K: PRINT
5060 ZL = A(1,5) + DZ * J
5070 FOR I = 0 TO NR
5080 RL = A(0,5) + DR * I: GOSUB 5400
5090 PRINT "ZL= ";ZL; TAB( 21);"RL= ";RL; TAB( 41);"ZVT= ";ZVT; TAB( 21);"RVT= ";RVT
5100 NEXT I
5105 PRINT : PRINT "DP= ";DP; TAB( 21);"HT= ";A(1,1)
5108 FOR K = 0 TO 39: PRINT "*";: NEXT K: PRINT : PRINT
5110 NEXT J
5120 PRINT R$;D$;"PR#0"
5130 RETURN : END
5200 REM VOLTAGE-->LOCATION ROUT
5210 DP = (ZVT - A(1,4)) * 1.3523:ZL = A(1,1) - DP + A(1,3):DP = DP - A(1,3)
5220 RA = 9.08 - RVT * 0.646:RL = RA + A(0,3)
5230 RETURN : END
5300 REM A(I,J)-->PL(I,J)
5310 PL(0,1) = A(0,1) / 2:PL(1,1) = A(1,1)
5320 PL(0,2) = 0:PL(1,2) = 0
5330 PL(0,3) = A(0,1) / 2:PL(1,3) = A(1,1)
5340 PL(0,4) = 0.5:PL(1,4) = 0.5
5350 RETURN : END
5400 REM GRID PT-->POTENTIO VLTG
5410 DP = A(1,1) - ZL + A(1,3):ZVT = A(1,4) + DP / 1.3523
5420 RA = RL - A(0,3):RVT = 9.08 - RA / 0.646
5430 RETURN : END
5500 INPUT "WANT TO PRINT OUT ? (Y/N) ";AN$: IF LEFT$(AN$,1) > < "Y" THEN GOTO 5520
5510 PRINT R$;D$;"PR#1": POKE 1401,80
5520 PRINT "ZL= ";ZL; TAB( 21);"RL= ";RL; TAB( 41);"ZVT= ";ZVT; TAB( 21);"RVT= ";RVT
5530 PRINT "DP= ";DP; TAB( 21);"HT= ";A(1,1)
5540 PRINT R$;D$;"PR#0"
5550 INPUT " KEEP ON ? (Y/N) ";AN$
5560 RETURN : END
6700 C = PEEK (116):D = PEEK (115): GOSUB 6900
6710 VTAB 3: PRINT "PRESENT HIMEM="; TAB( 33);A$
6720 VTAB 4: HTAB 32: PRINT "=";D + C * 256:Z1 = PEEK (37)
6730 VTAB 5: PRINT "WANT TO CHANGE HIMEM Y/N = ": VTAB 5: HTAB 28: INPUT AN$
6740 IF AN$ > < "Y" THEN GOTO 6780
6750 VTAB 5: HTAB 1: PRINT "INPUT NEW HIMEM "
6755 VTAB 5: HTAB 17: INVERSE : PRINT "(4-HEX;$XXXX)=": NORMAL : VTAB 5: HTAB 32: INPUT A$
6760 GOSUB 6800:B = C * 256 + D: VTAB 6: HTAB 32: PRINT "=";B
6770 HIMEM: B:Z1 = PEEK (37)
6780 POKE 34,Z1: HOME : RETURN : END
6800 FOR I = 2 TO 5
6810 A(0,I) = ASC ( MID$ ( A$,I,1))
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6820 IF A(O,I) = < 57 THEN A(O,I) = A(O,I) - 48: GOTO 6840
6830 A(O,I) = A(O,I) - 55
6840 NEXT I
6850 C = A(O,2) * 16 + A(O,3): D = A(O,4) * 16 + A(O,5)
6860 RETURN : END
6900 A(O,1) = INT (C / 16): A(O,2) = C - A(O,1) * 16: A(O,3) = INT (D / 16): A(O,4) = D - A(O,3) * 16: A$ = "$"
6910 FOR I = 1 TO 4
6920 IF A(O,I) < 10 THEN A$ = A$ + STR$ (A(O,I)): GOTO 6940
6930 A(O,I) = A(O,I) + 55: A$ = A$ + CHR$ (A(O,I))
6940 NEXT I
6950 RETURN : END
7000 REM SUBR FRAME (PL(I,J :I=0,1;J=0,1)
7010 B%(0) = INT (PL(O,0) * UPI) - 1: REM XREF
7020 B%(1) = INT (PL(1,0) * UPI) - 1: REM YREF
7030 B%(2) = INT ((PL(O,0) + PL(O,1)) * UPI) - 1: REM XLIM
7040 B%(3) = INT ((PL(1,0) + PL(1,1)) * UPI) - 1: REM YLIM
7050 FOR I = 0 TO 3
7060 POKE XREF + I * 2, INT (B%(I) / H): POKE XREF + I * 2 - 1, B%(I) - INT (B%(I) / H) * H
7070 NEXT I
7080 RETURN : END
7200 REM SUBR MKTIC (PL(I,J :I=0,1;J=0,1,2,3,4,5,6)
7202 FOR I = 0 TO 1
7204 J = 1 - I
7206 POKE TSFT + J, INT (PL(J,5) * UPI): POKE TLNG + J, H - INT (PL(J,6) * UPI)
7208 NEXT I
7210 FOR I = 0 TO 1
7220 IF PL(I,2) * PL(I,3) = > 0 THEN B%(I) = INT (PL(I,0) * UPI) - 1: GOTO 7240
7230 B%(I) = INT ((PL(I,0) - PL(I,2) * PL(I,1) / (PL(I,3) - PL(I,2))) * UPI) - 1
7240 POKE XZERO + I * 2, INT (B%(I) / H): POKE XZERO + I * 2 - 1, B%(I) - INT (B%(I) / H) * H
7250 J = 1 - I
7260 B%(I + 2) = INT ((PL(J,3) - PL(J,2)) / PL(J,4)): POKE TCNM + J, B%(I + 2)
7270 FOR K = 0 TO B%(I + 2)
7280 CR%(J,K) = INT ((PL(J,0) + PL(J,1) * PL(J,4) * K / ABS (PL(J,3) - PL(J,2))) * UPI) - 1
7290 NEXT K
7300 NEXT I
7310 B%(5) = PEEK (107) + PEEK (108) * H: REM PL(1,19) ARRAY ADDR
7320 B%(5) = B%(5) + 209 + 9 - 1: REM ADDR OF 1ST ELEMENT OF CR% - 1
7330 POKE CRARY, INT (B%(5) / H): POKE CRARY - 1, B%(5) - INT (B%(5) / H) * H
7340 RETURN : END
7400 REM SUBR CAPTIC (PL(I,J :I=0,1;J=0,2,3,4,7,8,9,10,11,12), CP$(J,0 :J=0,1))
7410 FOR I = 0 TO 1
7420 J = 1 - I
7430 B%(I + 2) = INT ((PL(J,3) - PL(J,2)) / PL(J,4) / PL(J,12))
7440 POKE TCNM + J, B%(I + 2): POKE RTT + J, PL(J,7)
7450 POKE WDT + J, PL(J,8)
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7460 POKE HGT + J,PL(J,9)
7470 NEXT I
7480 FOR I = 0 TO 1
7490 J = 1 - I
7500 IF PL(I,2) * PL(I,3) = > 0 THEN B%(I) = INT ((PL(I,0) - PL(J,10)) * UPI) - 1: GOTO 7520
7510 B%(I) = INT ((PL(I,0) - PL(I,2) * PL(I,1) / (PL(I,3) - PL(I,2)) - PL(J,10)) * UPI) - 1
7520 POKE XZERO + I * 2, INT (B%(I) / H): POKE XZERO + I * 2 - 1, B%(I) - INT (B%(I) / H) * H
7530 FOR K = 0 TO B%(I + 2)
7540 CR%(J,K) = INT ((PL(J,0) + PL(J,1) * PL(J,4) * PL(J,12) * K / ABS (PL(J,3) - PL(J,2)) - PL(J,11)) * UPI) - 1
7550 NEXT K
7560 NEXT I
7570 B%(5) = PEEK (107) + PEEK (108) * H: REM PL(1,19) ARRAY ADDR
7580 B%(5) = B%(5) + 209 + 9 - 1: REM ADDR OF 1ST ELEMENT OF CR% - 1
7590 POKE CRARY, INT (B%(5) / H): POKE CRARY - 1, B%(5) - INT (B%(5) / H) * H
7600 B%(6) = B%(5) + 169: REM ADDR OF 1ST ELEMENT OF CP$ - 1
7610 POKE CPARY, INT (B%(6) / H): POKE CPARY - 1, B%(6) - INT (B%(6) / H) * H
7620 RETURN : END
7700 REM SUBR CURVE/DOTPLT (PL(I,J : I=0,1;J=2,3,13,14,15,16),PL(1,17))
7702 POKE SYMBOL,PL(0,16): POKE RTT,PL(1,15)
7704 POKE WDT,PL(1,16): POKE HGT,PL(1,16)
7710 FOR K = 0 TO PL(0,15) - 1
7720 IF (CV%(0,K) / PL(0,13)) = > PL(0,2) THEN K1 = K: GOTO 7740
7730 NEXT K
7740 FOR K = PL(0,15) - 1 TO K1 STEP - 1
7750 IF (CV%(0,K) / PL(0,13)) = < PL(0,3) THEN K2 = K: GOTO 7770
7760 NEXT K
7770 B%(5) = K2 - K1 + 1
7780 POKE LPNMH, INT (B%(5) / H): POKE LPNMH - 1, B%(5) - INT (B%(5) / H) * H
7790 B%(0) = 2000: B%(1) = 1400: B%(2) = 1999: B%(3) = 1399
7800 FOR K = K1 TO K2
7810 FOR I = 0 TO 1
7820 CV%(I,K) = INT ((PL(I,0) + (CV%(I,K) / PL(I,13) - PL(I,2)) * PL(I,1) / (PL(I,3) - PL(I,2))) * UPI) - 1
7830 IF CV%(I,K) = > B%(I) THEN CV%(I,K) = B%(I + 2)
7840 NEXT I
7850 NEXT K
7860 B%(5) = PEEK (107) + PEEK (108) * H: REM PL(1,19) ARRAY ADDR
7870 B%(5) = B%(5) + 209 + 169 + 15 + 10 + 27 + 67 + 9: REM ADDR OF 1ST ELEMENT OF CV%
7880 B%(6) = B%(5) + 4 * K1 - 1: REM ADDR OF K1TH ELEMENT OF CV% - 1
7890 POKE CVARY, INT (B%(6) / H): POKE CVARY - 1, B%(6) - INT (B%(6) / H) * H
7900 RETURN : END
8000 REM COMMENT (R,W,HT,X,Y,CM$(0))
8010 POKE RTT,R: POKE WDT,W: POKE HGT,HT
8020 B%(0) = INT (X * UPI) - 1: B%(1) = INT (Y * UPI) - 1
8030 FOR I = 0 TO 1
8040 POKE XCRD + 2 * I, INT (B%(I) / H): POKE XCRD + 2 * I - 1, B%(I) - INT (B%(I) / H) * H
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```
8050 NEXT I
8060 B%(5) = PEEK (107) + PEEK (108) * H: REM PL(1,19) ARRAY ADDR
8070 B%(5) = B%(5) + 209 + 169 + 15 + 7 - 1: REM ADDR OF 1ST ELEMENT OF CM$-1
8080 POKE CMARY, INT (B%(5) / H): POKE CMARY - 1, B%(5) - INT (B%(5) / H) * H
8090 RETURN : END
8100 REM LINDRW (A(I, J: I=0, 1; J=11, 12), PL(I, J: I=0, 1; J=0, 1, 2, 3, 14))
8110 FOR K = 11 TO 12
8120 FOR I = 0 TO 1
8130 J = (K - 11) * 2 + I
8140 B%(J) = INT ((PL(I, 0) + (A(I, K) * PL(I, 14) - PL(I, 2)) * PL(I, 1) / (PL(I, 3) - PL(I, 2))) * UPI) - 1
8150 NEXT I
8160 NEXT K
8170 FOR I = 0 TO 3: POKE XCRD + I * 2, INT (B%(I) / H): POKE XCRD + I * 2 - 1, B%(I) - INT (B%(I) / H) * H: NEXT I
8180 RETURN : END
```



APPENDIX F**Mainframe Program Listings**



227

```
M=1
SIGN=1.
IF(I.EQ.1) M=3
IF(I.EQ.IN) M=5
IF(J.EQ.1) M=2
IF(J.EQ.JN) M=4
IF(J.EQ.J1.AND.I.LE.I1) M=4
IF(I.EQ.I1.AND.J.GT.J1) M=3
GO TO (1,2,3,4,5),M
C*** M=1, FOR POINTS NOT ON ANY OF THE BOUNDARIES
1 IF(LX.EQ.2) GO TO 11
PN=1.
BENQ=A(I+1,J,K)
BWSR=A(I-1,J,K)
BP=A(I,J,K)
XENQ=X1(I+1)-X1(I)
XWSR=X1(I)-X1(I-1)
GO TO 100
11 PN=1.
BENQ=A(I,J+1,K)
BWSR=A(I,J-1,K)
BP=A(I,J,K)
XENQ=X2(J+1)-X2(J)
XWSR=X2(J)-X2(J-1)
GO TO 100
C *** FOR POINTS ON THE SYMMETRY AXIS
2 IF(LX.EQ.1) GO TO 1
PN=-1.
BENQ=A(I,J+1,K)
BWSR=A(I,J+2,K)
BP=A(I,J,K)
XENQ=X2(J+1)-X2(J)
XWSR=X2(J+2)-X2(J)
GO TO 100
C*** M=3, FOR POINTS ON THE BOUNDARY I=I1
3 IF(LX.EQ.2) GO TO 11
PN=-1.
BENQ=A(I+1,J,K)
BWSR=A(I+2,J,K)
BP=A(I,J,K)
XENQ=X1(I+1)-X1(I)
XWSR=X1(I+2)-X1(I)
GO TO 100
C*** M=4, FOR POINTS ON THE BOUNDARY J=JN
4 IF(LX.EQ.1) GO TO 1
```

```
ADF00460
ADF00470
ADF00480
ADF00490
ADF00500
ADF00510
ADF00520
ADF00530
ADF00540
ADF00550
ADF00560
ADF00570
ADF00580
ADF00590
ADF00600
ADF00610
ADF00620
ADF00630
ADF00640
ADF00650
ADF00660
ADF00670
ADF00680
ADF00690
ADF00700
ADF00710
ADF00720
ADF00730
ADF00740
ADF00750
ADF00760
ADF00770
ADF00780
ADF00790
ADF00800
ADF00810
ADF00820
ADF00830
ADF00840
ADF00850
ADF00860
ADF00870
ADF00880
ADF00890
ADF00900
```

```
PN=-1.
BENQ=A(I,J-1,K)
BWSR=A(I,J-2,K)
BP=A(I,J,K)
XENQ=X2(J)-X2(J-1)
XWSR=X2(J)-X2(J-2)
SIGN=-1.
GO TO 100
C*** M=5, FOR POINTS ON THE BOUNDARY I=IN
5 IF(LX.EQ.2) GO TO 11
PN=-1.
BENQ=A(I-1,J,K)
BWSR=A(I-2,J,K)
BP=A(I,J,K)
XENQ=X1(I)-X1(I-1)
XWSR=X1(I)-X1(I-2)
SIGN=-1.
100 CONTINUE
ADF=((XENQ*XENQ-XWSR*XWSR)*BP+
1XWSR*XWSR*BENQ-XENQ*XENQ*BWSR)/(XENQ*XWSR*(PN*XENQ+XWSR))
ADF=SIGN*ADF
RETURN
END
```

ADFO0910  
ADFO0920  
ADFO0930  
ADFO0940  
ADFO0950  
ADFO0960  
ADFO0970  
ADFO0980  
ADFO0990  
ADFO1000  
ADFO1010  
ADFO1020  
ADFO1030  
ADFO1040  
ADFO1050  
ADFO1060  
ADFO1070  
ADFO1080  
ADFO1090  
ADFO1100  
ADFO1110  
ADFO1120  
ADFO1130

BLOCK DATA	BLK00010
C\$	BLK00020
COMMON/CNUMB/NW,NF,NK,NEP,NV1,NV2,NT,NMU,NRO,NTC,NSP,NMT	BLK00030
COMMON/CCORD/IMIN(21),IMAX(21),X1(31),X2(21),R(21)	BLK00040
COMMON/CGRID/IN,INM,JN,JNM,I1,I2,I3,IA,IW(21),J1	BLK00050
COMMON/CDIM/X(5),RE,RM	BLK00060
COMMON/CITER/NMAX,NPRINT,NITER,IP,IE,IV,CC	BLK00070
COMMON/CCONT/C1,C2,C3,CD	BLK00080
COMMON/CRLAX/RP(10),RSDU(10),SAN(10)	BLK00090
COMMON/CEMP/WF,P,S(4)	BLK00100
COMMON/CHJ/NHR,NHI,NJRR,NJRI,NJZR,NJZI,NJJ,NS	BLK00110
COMMON/CTHERM/TCREF(5),SPREF(5),BETA(2),ES,EE,EM,SB,HL	BLK00120
COMMON/CTEMP/TLM,TSM,TLS,TA,TM,TW	BLK00130
COMMON/CTRVL/VE,VC	BLK00140
COMMON/CHEAT/HC,HW(4)	BLK00150
COMMON/CPROP/ROREF(5),ZMUREF(2),PR(10),GAMA	BLK00160
COMMON/CVF/FES(21),FEM(21),FSM,FSE	BLK00170
COMMON/CDROP/TAU,HCD,D,QS,QM	BLK00180
COMMON/CTAUW/TAUW1(20),TAUW2(15),TAUW3(21),TAUW4(10),TAUW5(21)	BLK00190
1,CAPPA,E	BLK00200
COMMON/CFLD/CU	BLK00210
COMMON/CPI/PI	BLK00220
COMMON/CTYP/LTYP	BLK00230
COMMON/CANG/THETA	BLK00240
COMMON/FITER/MIT,MPRINT,CP	BLK00250
COMMON/CDVAR/A(31,21,20)	BLK00260
COMMON/CNAME/ANAME,ASYMBL	BLK00270
COMMON/CCRIT/NWI,NWJ,NFI,NFJ,NTI,NTJ,NKI,NKJ,NEI,NEJ,GOSA,TOTA	BLK00280
COMMON/CLF/LF,LI	BLK00290
COMMON/CSVOR/SVOR(31,21)	BLK00300
COMMON/CTURB/NTURB,NSCHEM,NVISC,RPXI,XI	BLK00310
COMMON/CSNW/SN1(20),SN2(15),SN3(21),SN4(20),SN5(21)	BLK00320
COMMON/CB/BE(31),BW(31),BN(21),BS(21)	BLK00330
COMMON/CAVM/AVM(60)	BLK00340
COMMON/CHBL/HLW,HLEH,HLEV,HLS,HLD,HLM,POWR	BLK00350
COMMON/CSUR/HS(2),TS(2)	BLK00360
COMMON/CBB/BBE,BBW,BBN,BBS,BPP	BLK00370
COMMON/CSORSE/SOURCE,SPRIME	BLK00380
COMMON/CGEOM/FR1,FR2(15),FR3,FR4,FR5	BLK00390
COMMON/CAF/FUN(20)	BLK00400
COMMON/CVOLT/VOLT	BLK00410
COMMON/CAVT/TB	BLK00420
COMMON/CYP/YP1(20),YP2(15),YP3(21),YP4(10),YP5(21)	BLK00430
C\$	BLK00440
COMMON/CADD/JA,JAX,RA,RAX,IAX	BLK00450

230

C .....	PROGRAM AND PRINT OUT CONTROL DATA	BLK00460
C .....	-----	BLK00470
C .....	DATA NW,NF,NK,NEP,NV1,NV2,NT,NMU,NRO,IE,IV/3,4,5,6,7,8,9,10,12,6,	BLK00480
C .....	16/	BLK00490
C .....	-----	BLK00500
C .....	DATA IP,CC/1,0.005/	BLK00510
C .....	2,RP/1.5,1.2,0.2,1.2,0.5,0.75,0.5,3*1.0/	BLK00520
C .....	3,NTC,NSP,NMT/13,14,15/	BLK00530
C .....	-----	BLK00540
C .....	DATA RS DU,SAN/10*0.0,10*0.0/	BLK00550
C .....	-----	BLK00560
C .....	DATA NHR,NHI,NJRR,NJRI,NJZR,NJZI,NJJ,NS /1,2,16,17,18,19,11,20/	BLK00570
C .....	1,MIT,MPRINT,CP/300,25.0.0001/	BLK00580
C .....	-----	BLK00590
C .....	PHYSICAL DATA	BLK00600
C .....	-----	BLK00610
C .....	DATA ROREF/5*08.37E+03/	BLK00620
C .....	1,ZMUREF/2*2.29E-02/	BLK00630
C .....	2,PR/4*1.0,1.000,1.3,4*1.0/	BLK00640
C .....	3,TCREF/7.0E-03,5.0E-03,7.0E-03,7.0E-03,7.0E-03/	BLK00650
C .....	4,SPREF,HL/0.16,0.31,0.16,1.1751,0.16,67.0/	BLK00660
C .....	5,TLM,TSM,TL,TA,TM,TW/300.0,300.0,300.0,300.0,300.0,300.0/	BLK00670
C .....	6,ES,EE,EM,SB/0.80,0.40,1.00,13.70/	BLK00680
C .....	7,BETA/2*0./	BLK00690
C .....	8,VE,VC,HC/0.,0.,5.0E-03/	BLK00700
C .....	9,GAMA/0.9/	BLK00710
C .....	1,HW/5.0E-02,5.0E-02,4.5E-02,5.0E-02/	BLK00720
C .....	2,CAPPA,E/0.40,9.0/	BLK00730
C .....	3,PI/3.1416/	BLK00740
C .....	DATA C1,C2,C3,CD/1.44,1.92,1.0,0.09/	BLK00750
C .....	DATA WF,P/0.0,1.26E-06/	BLK00760
C .....	1,S/5.84E+07,0.99E+06,0.99E+06,5.84E+07/	BLK00770
C .....	2,CU/0.50/	BLK00780
C .....	3,LTYP/2/	BLK00790
C .....	-----	BLK00800
C .....	GRID DATA	BLK00810
C .....	-----	BLK00820
C .....	DATA IN,JN/21,17/	BLK00830
C .....	1,X/0.01,0.010,0.0130,0.0425,0.050/	BLK00840
C .....	2,RE,RM,RA,RAX/0.0075,0.075,.0125,.0675/	BLK00850
C .....	3,I1,I2,IA,IAX,I3,J1,JA,JAX/7,7,10,18,21,5,08,15/	BLK00860
C .....	4,(IMIN(I),I=1,17)/05*2,12*8/	BLK00870
C .....	5,(IMAX(I),I=1,17)/17*20/	BLK00880
C .....	6,THETA/0.0000/	BLK00890
C .....	DATA TAU,HCD,D/0.,0.,0./	BLK00900

231

C \*\*\*\*\*  
END

E N D \*\*\*\*\*BLK00910  
BLK00920





233

```

          J11=J1-1
C..... AT THE CENTERLINE OF VESSEL
C-----
R3SQ =R(3)*R(3)
R2SQ =R(2)*R(2)
BB =R3SQ /(R3SQ -R2SQ )
DO 41 I=2,INM
      IF(I.LT.IW(1).OR.I.GT.I3) GO TO 41
A(I,1,NW)=(A(I,1,NF)-A(I,3,NF))/R3SQ+(A(I,2,NF)-A(I,1,NF))/R2SQ
A(I,1,NW) =8.*A(I,1,NW) / A(I,1,NRO)/(R3SQ-R2SQ)
      IF(NTURB.EQ.O) GO TO 41
A(I,1,NK)=BB*A(I,2,NK)+(1.-BB)*A(I,3,NK)
A(I,1,NEP)=BB*A(I,2,NEP)+(1.-BB)*A(I,3,NEP)
IF(A(I,1,NK).LE.O.O) A(I,J,NK)=O.O
IF(A(I,J,NEP).LE.O.O) A(I,J,NEP)=O.O
41      CONTINUE
C-----
C..... AT SIDE WALL
C-----
C..... VORTICITY
C-----
DX2=X2(JN)-X2(JNM)
      DO 42 I=I1,I3
Z=A(I,JN,NW)
A(I,JN,NW)=3.*(A(I,JN,NF)-A(I,JNM,NF))/DX2/DX2/R(JN)/A(I,JN,NRO)
A(I,JN,NW)=A(I,JN,NW)/R(JNM)-O.5*A(I,JNM,NW)
42      CONTINUE
C-----
C..... AT FREE SURFACE
C-----
          J12=J1+1
          DO 44 J=J12,JNM
A(I1,J,NW)=O.O
44      CONTINUE
C-----
C..... AT THE BASE ELECTRODE SURFACE
C-----
DX12=(X1(I3)-X1(I3-1))*2
      DO 43 J=2,JNM
RSQ=R(J)*R(J)
A(I3,J,NW)=3.*(A(I3,J,NF)-A(I3-1,J,NF))/(DX12*RSQ*RREF(2))
1-O.5*A(I3-1,J,NW)
43      CONTINUE
C-----
C..... ON THE TOP ELECTRODE SURFACE

```

BOU00460  
 BOU00470  
 BOU00480  
 BOU00490  
 BOU00500  
 BOU00510  
 BOU00520  
 BOU00530  
 BOU00540  
 BOU00550  
 BOU00560  
 BOU00570  
 BOU00580  
 BOU00590  
 BOU00600  
 BOU00610  
 BOU00620  
 BOU00630  
 BOU00640  
 BOU00650  
 BOU00660  
 BOU00670  
 BOU00680  
 BOU00690  
 BOU00700  
 BOU00710  
 BOU00720  
 BOU00730  
 BOU00740  
 BOU00750  
 BOU00760  
 BOU00770  
 BOU00780  
 BOU00790  
 BOU00800  
 BOU00810  
 BOU00820  
 BOU00830  
 BOU00840  
 BOU00850  
 BOU00860  
 BOU00870  
 BOU00880  
 BOU00890  
 BOU00900

234

```

C -----
      GFC=COS(THETA)
      DO 40 J=2,J1
      II=IW(J)
      XS=X2(J-1)/GFC+(X2(J)-X2(J-1))*GFC
      RSQ=(XS*GFC)**2
      DXN=(X1(II+1)-X1(II))*GFC
      DX12=DXN*DXN
      WW(J)=3.*(O.-A(II+1,J,NF))/DX12/RSQ/A(II,J,NRO)
      WW(J)=WW(J)-0.5*A(II+1,J,NW)
40      CONTINUE
      DO 48 J=2,J1
      II=IW(J)
      GF=SIN(THETA)/COS(THETA)
      DWF=GF/(GF+(X2(J+1)-X2(J))/(X1(II+1)-X1(II)))
      A(II,J,NW)=WW(J)
48      CONTINUE
C -----
C .....AT SIDE SURFACE OF TOP ELECTRODE(J=J1, I1<I<I2)
C -----
C (EXIT IF ELECTRODE JUST TOUCHES THE METAL SURFACE)
      IF(I1.EQ.I2) GO TO 47
      DX2=X2(J1+1)-X2(J1)
      DX22=DX2*DX2
      I21=I2-1
      RSQ=R(J1)*R(J1+1)
      XB=X2(J1)-X2(J11)
      DO 45 I=I1,I21
      Z=A(I,J1,NW)
      DT=A(I,J1,NT)-A(I,J1+1,NT)
      A(I,J1,NW)=3.*(A(I,J1,NF)-A(I,J1+1,NF))/DX22/RSQ/A(I,J1,NRO)
      A(I,J1,NW)=A(I,J1,NW)-0.5*A(I,J1+1,NW)
      A(I,J1,NW)=Z+RP(NW)*(A(I,J1,NW)-Z)
      TCM=TCREF(1)
45      CONTINUE
47      CONTINUE
C -----
      RETURN
      END

```

BOU00910  
BOU00920  
BOU00930  
BOU00940  
BOU00950  
BOU00960  
BOU00970  
BOU00980  
BOU00990  
BOU01000  
BOU01010  
BOU01020  
BOU01030  
BOU01040  
BOU01050  
BOU01060  
BOU01070  
BOU01080  
BOU01090  
BOU01100  
BOU01110  
BOU01120  
BOU01130  
BOU01140  
BOU01150  
BOU01160  
BOU01170  
BOU01180  
BOU01190  
BOU01200  
BOU01210  
BOU01220  
BOU01230  
BOU01240  
BOU01250  
BOU01260  
BOU01270  
BOU01280  
BOU01290

```
SUBROUTINE CONLOT(XI,PERFIN,PCASC)                                CON00010
C$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$ CON00020
COMMON/CNUMB/NW,NF,NK,NEP,NV1,NV2,NT,NMU,NRO,NTC,NSP,NMT        CON00030
COMMON/CCORD/IMIN(21),IMAX(21),X1(31),X2(21),R(21)            CON00040
COMMON/CGRID/IN,INM,JN,JNM,I1,I2,I3,IA,IW(21),J1              CON00050
COMMON/CDIM/X(5),RE,RM                                         CON00060
COMMON/CITER/NMAX,NPRINT,NITER,IP,IE,IV,CC                     CON00070
COMMON/CCONT/C1,C2,C3,CD                                        CON00080
COMMON/CRLAX/RP(10),RSDU(10),SAN(10)                            CON00090
COMMON/CEMP/WF,P,S(4)                                          CON00100
COMMON/CHJ/NHR,NHI,NJRR,NJRI,NJZR,NJZI,NJJ,NS                  CON00110
COMMON/CTEMP/TLM,TSM,TLS,TA,TM,TW                               CON00120
COMMON/CTRVL/VE,VC                                             CON00130
COMMON/CHEAT/HC,HW(4)                                          CON00140
COMMON/CPROP/ROREF(5),ZMUREF(2),PR(10),GAMA                    CON00150
COMMON/CVF/FES(21),FEM(21),FSM,FSE                             CON00160
COMMON/CDROP/TAU,HCD,D,QS,QM                                    CON00170
COMMON/CTAUW/TAUW1(20),TAUW2(15),TAUW3(21),TAUW4(10),TAUW5(21) CON00180
1,CAPPA,E                                                       CON00190
COMMON/CFLD/CU                                                  CON00200
COMMON/CPI/PI                                                   CON00210
COMMON/CTYP/LTYP                                               CON00220
COMMON/CANG/THETA                                              CON00230
COMMON/FITER/MIT,MPRINT,CP                                     CON00240
COMMON/CDVAR/A(31,21,20)                                       CON00250
COMMON/CNAME/ANAME,ASYMBL                                       CON00260
COMMON/CCRIT/NWI,NWJ,NFI,NFJ,NTI,NTJ,NKI,NKJ,NEI,NEJ,GOSA,TOTA CON00270
COMMON/CLF/LF,LI                                               CON00280
COMMON/CSVOR/SVOR(31,21)                                       CON00290
COMMON/CSNW/SN1(20),SN2(15),SN3(21),SN4(20),SN5(21)          CON00300
COMMON/CB/BE(31),BW(31),BN(21),BS(21)                          CON00310
COMMON/CAVM/AVM(60)                                             CON00320
COMMON/CHBL/HLW,HLEH,HLEV,HLS,HLD,HLM,POWR                    CON00330
COMMON/CSUR/HS(2),TS(2)                                         CON00340
COMMON/CBB/BBE,BBW,BBN,BBS,BPP                                  CON00350
COMMON/CSORSE/SOURCE,SPRIME                                       CON00360
COMMON/CGEOM/FR1,FR2(15),FR3,FR4,FR5                            CON00370
COMMON/CAF/FUN(20)                                              CON00380
COMMON/CVOLT/VOLT                                              CON00390
COMMON/CAVT/TB                                                  CON00400
COMMON/CYP/YP1(20),YP2(15),YP3(21),YP4(10),YP5(21)           CON00410
C$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$ CON00420
COMMON/CPROD/GK(31,21)                                          CON00430
COMMON/CSHEAR/GKK(31,31)                                        CON00440
DATA DIMNSN,XISET,ETA/O.04,50.0,O.50/                          CON00450
```

```

236
C
C-----
C ROUTINE TO CALCULATE LOTSIANSKII CONSTANT,XI
C OUTPUT FILE ON LOGICAL UNIT # 17
C-----
C
      IH=I3-1
      IL=I1+1
C... FIND THE POWER FLOW:
C-----
      PCASC=0.0
                DO 60 J=2,JNM
      DR=R(J+1)-R(J-1)
      IL1=IL+1
                DO 60 I=IL,IH
      DX=X1(I+1)-X1(I-1)
      DV=R(J)*DR*DX/4.0
      PCASC=PCASC-P*A(I,J,NV2)*R(J)*A(I,J,NHR)**2*1E6*DV
60      CONTINUE
C-----
C COMPUTE POWER LOST AS VISCOUS DISSIPATION AT WALL
C-----
      PWALL = 0.0
      CONST = 11.5/(RREF(1)**0.5)
C TOP ELECTRODE FACE:
C-----
                DO 350 J=2,J1
      DR= (R(J+1)-R(J-1))/2.0
      DAREA= DR*R(J)
      PWALL = PWALL +(ABS(TAUW2(J))**1.5)*DAREA*CONST
      YPLUS = SQRT(TAUW2(J)*RREF(1))*(X1(IL)-X1(I1))/ZMUREF(1)
      WRITE(25,999) YPLUS
350      CONTINUE
C BOTTOM ELECTRODE FACE:
C-----
                DO 360 J=2,JNM
      DR= (R(J+1)-R(J-1))/2.0
      DAREA= DR*R(J)
      PWALL = PWALL + (ABS(TAUW3(J))**1.5)*DAREA*CONST
      YPLUS = SQRT(TAUW3(J)*RREF(1))*(X1(IN)-X1(IH))/ZMUREF(1)
      WRITE(25,999) YPLUS
360      CONTINUE
C CYLINDRICAL WALL:
C-----
                DO 370 I=IL,IH
      N=I-IL+1

```

```

CON00460
CON00470
CON00480
CON00490
CON00500
CON00510
CON00520
CON00530
CON00540
CON00550
CON00560
CON00570
CON00580
CON00590
CON00600
CON00610
CON00620
CON00630
CON00640
CON00650
CON00660
CON00670
CON00680
CON00690
CON00700
CON00710
CON00720
CON00730
CON00740
CON00750
CON00760
CON00770
CON00780
CON00790
CON00800
CON00810
CON00820
CON00830
CON00840
CON00850
CON00860
CON00870
CON00880
CON00890
CON00900

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          DX= (X1(I+1)-X1(I-1))/2.0          CON00910
          DAREA = RM*DX                      CON00920
          PWALL = PWALL + (ABS(TAUW1(N))*1.5)*DAREA*CONST CON00930
          YPLUS = SQRT(TAUW1(N)*ROREF(1))*(R(JN)-R(JNM))/ZMUREF(1) CON00940
          WRITE(25,999) YPLUS                CON00950
999      FORMAT(' YPLUS = ',1PE10.2)        CON00960
370      CONTINUE                            CON00970
C      EFFICIENCY OF ENERGY CONVERSION    CON00980
C      -----                              CON00990
          ETA2 = (PCASC-PWALL)/PCASC         CON01000
          WRITE(6,400) ETA2,PWALL,PCASC     CON01010
400      FORMAT('COMPUTED EFFICIENCY = ',F5.2 CON01020
1/'VISCOUS DISSIPATION = ',1PE10.2        CON01030
2/'TOTAL POWER FLOW   = ',1PE10.2)        CON01040
C-----CON01050
          PINPT=ETA*PCASC                   CON01060
          CALL VOL(GKK,SHEAR2)              CON01070
          IF(SHEAR2.EQ.0.0) GO TO 100        CON01080
          AVGVIS=PINPT/(SHEAR2*ROREF(1))    CON01090
          AI   =AVGVIS**2*DIMNSN**3        CON01100
          XI   =1.0/(AI**0.118)            CON01110
          PERFIN =AVGVIS*ROREF(1)/ZMUREF(1) CON01120
C-----CON01130
          RETURN                             CON01140
100      XI=XISET                           CON01150
          RETURN                             CON01160
          END                                CON01170

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SUBROUTINE CONVEC(AE,AW,AN,AS,I,J,K) CON00010
C$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$ CON00020
COMMON/CNUMB/NW,NF,NK,NEP,NV1,NV2,NT,NMU,NRO,NTC,NSP,NMT CON00030
COMMON/CCORD/IMIN(21),IMAX(21),X1(31),X2(21),R(21) CON00040
COMMON/CGRID/IN,INM,JN,JNM,I1,I2,I3,IA,IW(21),J1 CON00050
COMMON/CDIM/X(5),RE,RM CON00060
COMMON/CITER/NMAX,NPRINT,NITER,IP,IE,IV,CC CON00070
COMMON/CCONT/C1,C2,C3,CD CON00080
COMMON/CRLAX/RP(10),RSDU(10),SAN(10) CON00090
COMMON/CEMP/WF,P,S(4) CON00100
COMMON/CHJ/NHR,NHI,NJRR,NJRI,NJZR,NJZI,NJU,NS CON00110
COMMON/CTHERM/TCREF(5),SPREF(5),BETA(2),ES,EE,EM,SB,HL CON00120
COMMON/CTEMP/TLM,TSM,TLS,TA,TM,TW CON00130
COMMON/CTRVL/VE,VC CON00140
COMMON/CHEAT/HC,HW(4) CON00150
COMMON/CPROP/ROREF(5),ZMUREF(2),PR(10),GAMA CON00160
COMMON/CVF/FES(21),FEM(21),FSM,FSE CON00170
COMMON/CDROP/TAU,HCD,D,QS,QM CON00180
COMMON/CTAUW/TAUW1(20),TAUW2(15),TAUW3(21),TAUW4(10),TAUW5(21) CON00190
1,CAPPA,E CON00200
COMMON/CFLD/CU CON00210
COMMON/CPI/PI CON00220
COMMON/CTYP/LTYP CON00230
COMMON/CANG/THETA CON00240
COMMON/FITER/MIT,MPRINT,CP CON00250
COMMON/CDVAR/A(31,21,20) CON00260
COMMON/CNAME/ANAME,ASYMBL CON00270
COMMON/CCRIT/NWI,NWJ,NFI,NFJ,NTI,NTJ,NKI,NKJ,NEI,NEJ,GOSA,TOTA CON00280
COMMON/CLF/LF,LI CON00290
COMMON/CSVOR/SVOR(31,21) CON00300
COMMON/CTURB/NTURB,NSCHEM,NVISC,RPXI,XI CON00310
COMMON/CSNW/SN1(20),SN2(15),SN3(21),SN4(20),SN5(21) CON00320
COMMON/CB/BE(31),BW(31),BN(21),BS(21) CON00330
COMMON/CAVM/AVM(60) CON00340
COMMON/CHBL/HLW,HLEH,HLEV,HLS,HLD,HLM,POWR CON00350
COMMON/CSUR/HS(2),TS(2) CON00360
COMMON/CBB/BBE,BBW,BBN,BBS,BPP CON00370
COMMON/CSORSE/SOURCE,SPRIME CON00380
COMMON/CGEOM/FR1,FR2(15),FR3,FR4,FR5 CON00390
COMMON/CAF/FUN(20) CON00400
COMMON/CVOLT/VOLT CON00410
COMMON/CAVT/TB CON00420
COMMON/CYP/YP1(20),YP2(15),YP3(21),YP4(10),YP5(21) CON00430
C$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$$ CON00440
COMMON/CPROD/GK(31,21) CON00450

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COMMON/CSHEAR/GKK(31,31)
COMMON/CLREYN/LREYN,CDMOD(31,31)
C.....ROUTINE FOR CALCULATION OF AE,AW,AN,AS
C-----
C
C... CALCULATION MEAN MASS FLOW RATE AT FOUR TUBES OF THE TANK
C-----
DV=R(J)*(X1(I+1)-X1(I-1))*(X2(J+1)-X2(J-1))
G1PW=(A(I,J+1,NF)-A(I,J-1,NF)+A(I-1,J+1,NF)-A(I-1,J-1,NF))/DV
G1PE=(A(I,J+1,NF)-A(I,J-1,NF)+A(I+1,J+1,NF)-A(I+1,J-1,NF))/DV
G2PS=(A(I-1,J,NF)-A(I+1,J,NF)+A(I-1,J-1,NF)-A(I+1,J-1,NF))/DV
G2PN=(A(I-1,J,NF)-A(I+1,J,NF)+A(I-1,J+1,NF)-A(I+1,J+1,NF))/DV
C-----
C.....ACCOUNT FOR THE WALL FUNCTION
C-----
IF(K.EQ.NHR.OR.K.EQ.NHI) GO TO 10
IF(K.EQ.NW) GO TO 10
IF(J.NE.JNM) GO TO 7
C
C    CYLINDRICAL WALL NEIGHBOR POINTS
C-----
G1PE=- (A(I,J-1,NF)+A(I,J,NF)+A(I+1,J,NF)+A(I+1,J-1,NF))/DV
G1PW=- (A(I,J-1,NF)+A(I,J,NF)+A(I-1,J,NF)+A(I-1,J-1,NF))/DV
BBE=BBE*FR1
BBW=BBW*FR1
BBN=0.0
G2PN=0.
TAUW=TAUW1(I-I1)
IF(K.NE.NK.AND.K.NE.NEP) GO TO 16
DX=X2(JN)-X2(JNM-1)
UA=ABS(A(I,JNM,NV1)+A(I,JNM-1,NV1))/DX
YP=X2(JN)-X2(JNM)
IF(K.EQ.NK) GO TO 20
IF(K.EQ.NEP) GO TO 21
C-----
16 B1=FR1
B2=2.*R(J+1)/R(J)/(X2(J+1)-X2(J-1))
IF(K.NE.NT) GO TO 160
SNW=SN1(I-I1)
TWALL=A(I,J+1,NT)
GO TO 22
160 CONTINUE
GO TO 24
C-----
C    LOWER SURFACE OF TOP ELECTRODE

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CON00460
CON00470
CON00480
CON00490
CON00500
CON00510
CON00520
CON00530
CON00540
CON00550
CON00560
CON00570
CON00580
CON00590
CON00600
CON00610
CON00620
CON00630
CON00640
CON00650
CON00660
CON00670
CON00680
CON00690
CON00700
CON00710
CON00720
CON00730
CON00740
CON00750
CON00760
CON00770
CON00780
CON00790
CON00800
CON00810
CON00820
CON00830
CON00840
CON00850
CON00860
CON00870
CON00880
CON00890
CON00900

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C
7
-----
      IF(I.NE.(IW(J)+1)) GO TO 8
      IF(J.GT.J1) GO TO 8
GF=SIN(THETA)/COS(THETA)
G2PN=-(A(I,J+1,NF)+A(I,J,NF)+A(I+1,J,NF)+A(I+1,J+1,NF))/DV
G2PS=-(A(I,J-1,NF)+A(I,J,NF)+A(I+1,J,NF)+A(I+1,J-1,NF))/DV
G1PW=0.
FRN=FR2(J)+(X2(J+1)-X2(J))/(X1(I+1)-X1(I))*GF
FRS=FR2(J)-(X2(J)-X2(J-1))/(X1(I+1)-X1(I))*GF
BBN=BBN*FRN
BBS=BBS*FRS
BBW=0.0
TAUW=TAUW2(J)
      IF(K.NE.NK.AND.K.NE.NEP) GO TO 18
      DX=X1(I+1)-X1(I-1)
UA1=A(I,J,NV2)-A(I,J,NV1)*GF
UA2=A(I+1,J,NV2)-A(I,J,NV1)*GF
UA=ABS(UA1+UA2)/DX
YP=(X1(I)-X1(I-1))*COS(THETA)
      IF(K.EQ.NK) GO TO 20
      IF(K.EQ.NEP) GO TO 21
C-----
18 B1=FR2(J)
   B2=2./(X1(I+1)-X1(I-1))
   IF(K.NE.NT) GO TO 180
   SNW=SN2(J)
   TWALL=A(I-1,J,NT)
      GO TO 22
180 CONTINUE
   GO TO 24
C-----
C
C
C
8
-----
      IF(I.NE.(I3-1)) GO TO 9
G2PN= (A(I,J+1,NF)+A(I,J,NF)+A(I-1,J,NF)+A(I-1,J+1,NF))/DV
G2PS= (A(I,J-1,NF)+A(I,J,NF)+A(I-1,J,NF)+A(I-1,J-1,NF))/DV
G1PE=0.
BBE=0.0
BBN=BBN*FR3
BBS=BBS*FR3
TAUW=TAUW3(J)
      IF(K.NE.NK.AND.K.NE.NEP) GO TO 26
      DX=X1(I+1)-X1(I-1)
UA=ABS(A(I3-1,J,NV2)+A(I3-2,J,NV2))/DX
YP=X1(I+1)-X1(I)
CONO0910
CONO0920
CONO0930
CONO0940
CONO0950
CONO0960
CONO0970
CONO0980
CONO0990
CONO1000
CONO1010
CONO1020
CONO1030
CONO1040
CONO1050
CONO1060
CONO1070
CONO1080
CONO1090
CONO1100
CONO1110
CONO1120
CONO1130
CONO1140
CONO1150
CONO1160
CONO1170
CONO1180
CONO1190
CONO1200
CONO1210
CONO1220
CONO1230
CONO1240
CONO1250
CONO1260
CONO1270
CONO1280
CONO1290
CONO1300
CONO1310
CONO1320
CONO1330
CONO1340
CONO1350

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                IF(K.EQ.NK) GO TO 20          CONO1360
                IF(K.EQ.NEP) GO TO 21        CONO1370
C-----
26  B1=FR3                                     CONO1380
    B2=2./(X1(I+1)-X1(I-1))                 CONO1390
        IF(K.NE.NT) GO TO 261              CONO1400
    SNW=SN3(J)                               CONO1410
    TWALL=A(I+1,J,NT)                       CONO1420
        GO TO 22                           CONO1430
261  CONTINUE                                CONO1440
        GO TO 24                            CONO1450
C-----
C          SIDE SURFACE OF TOP ELECTRODE    CONO1460
C-----
9    IF(J.NE.(J1+1)) GO TO 10              CONO1470
    IF(I.GT.I2) GO TO 10                   CONO1480
G1PE= (A(I,J+1,NF)+A(I,J,NF)+A(I+1,J,NF)+A(I+1,J+1,NF))/DV CONO1490
G1PW= (A(I,J+1,NF)+A(I,J,NF)+A(I-1,J,NF)+A(I-1,J+1,NF))/DV CONO1500
G2PS=0.                                    CONO1510
BBS=0.                                     CONO1520
BBE=BBE*FR4                               CONO1530
BBW=BBW*FR4                               CONO1540
TAUW=TAUW4(I-I1)                          CONO1550
        IF(K.NE.NK.AND.K.NE.NEP) GO TO 28 CONO1560
    DX=X2(J1+2)-X2(J1)                     CONO1570
    UA=ABS(A(I,J1+1,NV1)+A(I,J1+2,NV1))/DX CONO1580
    YP=X2(J1+1)-X2(J1)                     CONO1590
        IF(K.EQ.NK) GO TO 20               CONO1600
        IF(K.EQ.NEP) GO TO 21             CONO1610
C-----
28  B1=FR4                                     CONO1620
    B2=2.*R(J-1)/R(J)/(X2(J+1)-X2(J-1))    CONO1630
        IF(K.NE.NT) GO TO 280            CONO1640
    SNW=SN4(I-I1)                           CONO1650
    TWALL=A(I,J-1,NT)                       CONO1660
        GO TO 22                           CONO1670
280  CONTINUE                                CONO1680
        GO TO 24                            CONO1690
C-----
20  SOURCE=UA*TAUW                           CONO1700
    SPRIME=UA*CDMOD(I,J)*A(I,J,NRO)**2*A(I,J,NK)/TAUW CONO1710
C    SPRIME=A(I,J,NEP)*A(I,J,NRO)/A(I,J,NK) CONO1720
        GK(I,J)=SOURCE                    CONO1730
        GKK(I,J)=UA**2                     CONO1740
        GO TO 10                           CONO1750
    CONO1760
    CONO1770
    CONO1780
    CONO1790
    CONO1800

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21 SOURCE      =(CDMOD(I,J)**0.75)*(A(I,J,NK)**1.50)/CAPPA/YP      CONO1810
C 21 SOURCE      =UA*CDMOD(I,J)*A(I,J,NRO)*A(I,J,NK)**2/TAUW      CONO1820
    SPRIME=1.0      CONO1830
    AE=0.          CONO1840
    AW=0.          CONO1850
    AN=0.          CONO1860
    AS=0.          CONO1870
    BBE=0.         CONO1880
    BBW=0.         CONO1890
    BBN=0.         CONO1900
    BBS=0.         CONO1910
                    IF(NSCHEM.EQ.1) A(I,J,NEP)=SOURCE      CONO1920
                    GO TO 100      CONO1930
22 SOURCE=SOURCE*B1+B2*SNW*TWALL-B3      CONO1940
    SPRIME=SNW*B2      CONO1950
                    GO TO 10      CONO1960
24 SOURCE=SOURCE*B1      CONO1970
10      CONTINUE      CONO1980
-----
C      COMPUTE THE CONVECTION CONSTANTS      CONO1990
C      -----      CONO2000
C      APP=1.      CONO2010
    IF(K.EQ.NHR.OR.K.EQ.NHI) APP=P/A(I,J,NRO)      CONO2020
    IF(K.EQ.NF) APP=0.      CONO2030
    IF(K.EQ.NW) APP=R(J)*R(J)      CONO2040
    IF(K.EQ.NT) APP=A(I,J,NSP)      CONO2050
    AW=0.5*APP*(ABS(G1PW)+G1PW)      CONO2060
    AE=0.5*APP*(ABS(G1PE)-G1PE)      CONO2070
    AN=0.5*APP*(ABS(G2PN)-G2PN)      CONO2080
    AS=0.5*APP*(ABS(G2PS)+G2PS)      CONO2090
100      CONTINUE      CONO2100
C#####          END #####      CONO2110
    RETURN      CONO2120
    END      CONO2130
          CONO2140

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SUBROUTINE CORD	COR00010
C\$	COR00020
COMMON/CNUMB/NW,NF,NK,NEP,NV1,NV2,NT,NMU,NRO,NTC,NSP,NMT	COR00030
COMMON/CCORD/IMIN(21),IMAX(21),X1(31),X2(21),R(21)	COR00040
COMMON/CGRID/IN,INM,JN,JNM,I1,I2,I3,IA,IW(21),J1	COR00050
COMMON/CDIM/X(5),RE,RM	COR00060
COMMON/CITER/NMAX,NPRINT,NITER,IP,IE,IV,CC	COR00070
COMMON/CCONT/C1,C2,C3,CD	COR00080
COMMON/CRLAX/RP(10),RSDU(10),SAN(10)	COR00090
COMMON/CEMP/WF,P,S(4)	COR00100
COMMON/CHJ/NHR,NHI,NJRR,NJRI,NJZR,NJZI,NJJ,NS	COR00110
COMMON/CTHERM/TCREF(5),SPREF(5),BETA(2),ES,EE,EM,SB,HL	COR00120
COMMON/CTEMP/TLM,TSM,TLS,TA,TM,TW	COR00130
COMMON/CTRVL/VE,VC	COR00140
COMMON/CHEAT/HC,HW(4)	COR00150
COMMON/CPROP/ROREF(5),ZMUREF(2),PR(10),GAMA	COR00160
COMMON/CVF/FES(21),FEM(21),FSM,FSE	COR00170
COMMON/CDROP/TAU,HCD,D,QS,QM	COR00180
COMMON/CTAUW/TAUW1(20),TAUW2(15),TAUW3(21),TAUW4(10),TAUW5(21)	COR00190
1,CAPPA,E	COR00200
COMMON/CFLD/CU	COR00210
COMMON/CP1/PI	COR00220
COMMON/CTYP/LTYP	COR00230
COMMON/CANG/THETA	COR00240
COMMON/FITER/MIT,MPRINT,CP	COR00250
COMMON/CDVAR/A(31,21,20)	COR00260
COMMON/CNAME/ANAME,ASYMBL	COR00270
COMMON/CCRIT/NWI,NWJ,NFI,NFJ,NTI,NTJ,NKI,NKJ,NEI,NEJ,GOSA,TOTA	COR00280
COMMON/CLF/LF,LI	COR00290
COMMON/CSVOR/SVOR(31,21)	COR00300
COMMON/CTURB/NTURB,NSCHEM,NVISC,RPXI,XI	COR00310
COMMON/CSNW/SN1(20),SN2(15),SN3(21),SN4(20),SN5(21)	COR00320
COMMON/CB/BE(31),BW(31),BN(21),BS(21)	COR00330
COMMON/CAVM/AVM(60)	COR00340
COMMON/CHBL/HLW,HLEH,HLEV,HLS,HLD,HLM,POWR	COR00350
COMMON/CSUR/HS(2),TS(2)	COR00360
COMMON/CBB/BBE,BBW,BBN,BBS,BPP	COR00370
COMMON/CSORSE/SOURCE,SPRIME	COR00380
COMMON/CGEOM/FR1,FR2(15),FR3,FR4,FR5	COR00390
COMMON/CAF/FUN(20)	COR00400
COMMON/CVOLT/VOLT	COR00410
COMMON/CAVT/TB	COR00420
COMMON/CYP/YP1(20),YP2(15),YP3(21),YP4(10),YP5(21)	COR00430
C\$	COR00440
COMMON/CADD/JA,JAX,RA,RAX,IAX	COR00450

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C          GRID GENERATION SECTION
C          -----
C          X1(1)=0.0
C          X2(1)=0.0
C          R(1)=X2(1)
C          IM=I2+J1-1
C          IF(THETA.EQ.O.) IM=I2
C          IW(J1)=I2
C          IW(1)=IM
C          GF=SIN(THETA)/COS(THETA)
C          XIM=X(2)+RE*GF
C          DO 11 J=2,J1
C          IW(J)=IW(J-1)-1
C          IF(J.GE.J1) IW(J)=I2
C          IF(THETA.EQ.O.) IW(J)=I2
11          CONTINUE
C          DZ1=X(1)/FLOAT(I1-1)
C          DZ3=(X(3)-X(2))/FLOAT(IA-I2)
C          DZ4=(X(4)-X(3))/FLOAT(IA-IA)
C          DZ5=(X(5)-X(4))/FLOAT(IN-IA)
C          DO 10 I=2,IN
C          IF(I.LE.I1) DX1=DZ1
C          IF(I.GT.I1.AND.I.LE.I2) DX1=DZ2
C          IF(I.GT.I2.AND.I.LE.IA) DX1=DZ3
C          IF(I.GT.IA.AND.I.LE.IAX) DX1=DZ4
C          IF(I.GT.IAX.AND.I.LE.IN) DX1=DZ5
10          X1(I)=X1(I-1)+DX1
C          DR1=RE/FLOAT(J1-1)
C          DR2=(RA-RE)/FLOAT(JA-J1)
C          DR3=(RAX-RA)/FLOAT(JAX-JA)
C          DR4=(RM-RAX)/FLOAT(JN-JAX)
C          DO 50 J=2,JN
C          DX2=DR1
C          IF(J.GT.J1.AND.J.LE.JA) DX2=DR2
C          IF(J.GT.JA.AND.J.LE.JAX) DX2=DR3
C          IF(J.GT.JAX.AND.J.LE.JN) DX2=DR4
C          X2(J)=X2(J-1)+DX2
C          R(J)=X2(J)
50          CONTINUE
C          -----
C          CALCULATE GRID BASED CONSTANTS
C          -----
C          DO 20 I=2,INM
C          DX1=1./(X1(I+1)-X1(I-1))

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COR00460
COR00470
COR00480
COR00490
COR00500
COR00510
COR00520
COR00530
COR00540
COR00550
COR00560
COR00570
COR00580
COR00590
COR00600
COR00610
COR00620
COR00630
COR00640
COR00650
COR00660
COR00670
COR00680
COR00690
COR00700
COR00710
COR00720
COR00730
COR00740
COR00750
COR00760
COR00770
COR00780
COR00790
COR00800
COR00810
COR00820
COR00830
COR00840
COR00850
COR00860
COR00870
COR00880
COR00890
COR00900

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	BW(I)=DX1/(X1(I)-X1(I-1))	COR00910
	BE(I)=DX1/(X1(I+1)-X1(I))	COR00920
20	CONTINUE	COR00930
	DO 21 J=2, JNM	COR00940
	DX2=1.0/(X2(J+1)-X2(J-1))	COR00950
	A1=DX2/(X2(J+1)-X2(J))	COR00960
	BN(J)=(1.+R(J+1)/R(J))*0.5*A1	COR00970
	A2=DX2/(X2(J)-X2(J-1))	COR00980
	BS(J)=(1.+R(J-1)/R(J))*0.5*A2	COR00990
21	CONTINUE	COR01000
23	CONTINUE	COR01010
	FR1=2.-(X2(JNM)-X2(JNM-1))/(X2(JN)-X2(JNM-1))	COR01020
	FR3=2.-(X1(I3-1)-X1(I3-2))/(X1(I3)-X1(I3-2))	COR01030
	FR4=2.-(X2(J1+2)-X2(J1+1))/(X2(J1+2)-X2(J1))	COR01040
	FR5=2.-(X1(I1+2)-X1(I1+1))/(X1(I1+2)-X1(I1))	COR01050
	DO 25 J=1, J1	COR01060
	II=IW(J)	COR01070
25	FR2(J)=2.-(X1(II+2)-X1(II+1))/(X1(II+2)-X1(II))	COR01080
	RETURN	COR01090
	END	COR01100

SUBROUTINE DISSIP	DIS00010
C\$	DIS00020
COMMON/CNUMB/NW,NF,NK,NEP,NV1,NV2,NT,NMU,NRD,NTC,NSP,NMT	DIS00030
COMMON/CCORD/IMIN(21),IMAX(21),X1(31),X2(21),R(21)	DIS00040
COMMON/CGRID/IN,INM,JN,JNM,I1,I2,I3,IA,IW(21),J1	DIS00050
COMMON/CDIM/X(5),RE,RM	DIS00060
COMMON/CITER/NMAX,NPRINT,NITER,IP,IE,IV,CC	DIS00070
COMMON/CCONT/C1,C2,C3,CD	DIS00080
COMMON/CRLAX/RP(10),RSDU(10),SAN(10)	DIS00090
COMMON/CEMP/WF,P,S(4)	DIS00100
COMMON/CHJ/NHR,NHI,NJRR,NJRI,NJZR,NJZI,NJJ,NS	DIS00110
COMMON/CTHERM/TCREF(5),SPREF(5),BETA(2),ES,EE,EM,SB,HL	DIS00120
COMMON/CTEMP/TLM,TSM,TLS,TA,TM,TW	DIS00130
COMMON/CTRVL/VE,VC	DIS00140
COMMON/CHEAT/HC,HW(4)	DIS00150
COMMON/CPROP/ROREF(5),ZMUREF(2),PR(10),GAMA	DIS00160
COMMON/CVF/FES(21),FEM(21),FSM,FSE	DIS00170
COMMON/CDROP/TAU,HCD,D,QS,QM	DIS00180
COMMON/CTAUW/TAUW1(20),TAUW2(15),TAUW3(21),TAUW4(10),TAUW5(21)	DIS00190
1,CAPPA,E	DIS00200
COMMON/CFLD/CU	DIS00210
COMMON/CPI/PI	DIS00220
COMMON/CANG/THETA	DIS00230
COMMON/FITER/MIT,MPRINT,CP	DIS00240
COMMON/CDVAR/A(31,21,20)	DIS00250
COMMON/CNAME/ANAME,ASYMBL	DIS00260
COMMON/CCRIT/NWI,NWJ,NFI,NFJ,NTI,NTJ,NKI,NKJ,NEI,NEJ,GOSA,TOTA	DIS00270
COMMON/CLF/LF,LI	DIS00280
COMMON/CTYP/LTYP	DIS00290
COMMON/CSVOR/SVOR(31,21)	DIS00300
COMMON/CTURB/NTURB,NSCHEM,NVISC,RPXI,XI	DIS00310
COMMON/CSNW/SN1(20),SN2(15)	DIS00320
COMMON/CB/BE(31),BW(31),BN(21),BS(21)	DIS00330
COMMON/CAVM/AVM(60)	DIS00340
COMMON/CHBL/HLW,HLEH,HLEV,HLS,HLD,HLM,POWR	DIS00350
COMMON/CSUR/HS(2),TS(2)	DIS00360
COMMON/CBB/BBE,BBW,BBN,BBS,BPP	DIS00370
COMMON/CSORSE/SOURCE,SPRIME	DIS00380
COMMON/CGEOM/FR1,FR2(15),FR3,FR4,FR5	DIS00390
COMMON/CAF/FUN(20)	DIS00400
COMMON/CVOLT/VOLT	DIS00410
COMMON/CAVT/TB	DIS00420
COMMON/CYP/YP1(20),YP2(15),YP3(21),YP4(10),YP5(21)	DIS00430
C\$	DIS00440
DATA IVAR/O/	DIS00450

247

C	.....CALCULATE DISSIPATION ENERGY	DIS00460
C	-----	DIS00470
	DO 31 J=1,JNM-1	DIS00480
	IL=I1+1	DIS00490
	ITOP=IL	DIS00500
	IF(J.LE.J1) ITOP=IL+1	DIS00510
	IH=I3-1	DIS00520
	DO 31 I=ITOP,IH-1	DIS00530
	IF(A(I,J,NK).LE.O.O) GO TO 41	DIS00540
	A(I,J,NEP)=(A(I,J,NK)*XI)**1.7	DIS00550
	GO TO 31	DIS00560
41	A(I,J,NK)=O.O	DIS00570
	A(I,J,NEP)=O.O	DIS00580
31	CONTINUE	DIS00590
C	GET DISSIPATION AT THE BOUNDARIES	DIS00600
C	-----	DIS00610
	DO 50 J=2,JNM	DIS00620
	IL=I1+1	DIS00630
	IH=I3-1	DIS00640
	DO 50 I=IL,IH	DIS00650
C	TOP SURFACE	DIS00660
C	-----	DIS00670
	IF(I.EQ.IL.AND.J.LE.J1) CALL CONVEC(D1,D2,D3,D4,I,J,NEP)	DIS00680
C	BOTTOM SURFACE	DIS00690
C	-----	DIS00700
	IF(I.EQ.IH) CALL CONVEC(D1,D2,D3,D4,I,J,NEP)	DIS00710
C	CYLINDRICAL SURFACE	DIS00720
C	-----	DIS00730
	IF(J.EQ.JNM) CALL CONVEC(D1,D2,D3,D4,I,J,NEP)	DIS00740
50	CONTINUE	DIS00750
	RETURN	DIS00760
	END	DIS00770







250

C		EMFO0910
C.....	COMPUTE AND PRINT COORDINATES	EMFO0920
C	-----	EMFO0930
345	CONTINUE	EMFO0940
	CALL CORD	EMFO0950
	WRITE(6,9)	EMFO0960
	WRITE(6,12) I1,I2,I3,IA,IN,J1,JN	EMFO0970
	WRITE(6,101) (X1(I),I=1,IN)	EMFO0980
	WRITE(6,102) (X2(J),J=1,JN)	EMFO0990
	WRITE(6,40)	EMFO1000
350	CONTINUE	EMFO1010
C.....	INITIALIZE	EMFO1020
C	-----	EMFO1030
	CALL INIT	EMFO1040
	LV=NHR	EMFO1050
	IF(INREAD.EQ.1)GO TO 989	EMFO1060
C.....	IF READ IN NOT DESIRED THEN USE THE VALUES GENERATED BY I N I T	EMFO1070
	GO TO 999	EMFO1080
989	CONTINUE	EMFO1090
C.....	READ IN SEGMENT; UNIT NUMBERS 4 AND 8	EMFO1100
C	-----	EMFO1110
	DO 21 K=NHR,LV	EMFO1120
	DO 21 J=1,JN	EMFO1130
21	READ(4,14) (A(I,J,K),I=1,IN)	EMFO1140
	DO 22 K=NW,IE	EMFO1150
	DO 22 J=1,JN	EMFO1160
22	READ(8,14) (A(I,J,K),I=I1,I3)	EMFO1170
999	CONTINUE	EMFO1180
23	CONTINUE	EMFO1190
	LI=1	EMFO1200
	CALL PROP(1)	EMFO1210
C.....	SOLVE MAXWELL'S EQUATIONS	EMFO1220
C	-----	EMFO1230
	IF(LI.EQ.2) GO TO 100	EMFO1240
	IF(NFIELD.EQ.1)CALL FIELD	EMFO1250
979	CONTINUE	EMFO1260
	IF(LF.EQ.2) GO TO 30	EMFO1270
100	CONTINUE	EMFO1280
C.....	PERFORM FLUID FLOW CALCULATIONS	EMFO1290
C	-----	EMFO1300
	NITER=0	EMFO1310
1	CONTINUE	EMFO1320
C	=====	EMFO1330
	NITER=NITER+1	EMFO1340
C.....	CAUSE ONE CYCLE OF ITERATION TO BE PERFORMED	EMFO1350

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C                                     IF(NFLOW.EQ.0) GO TO 30
CALL EQN
WRITE(15,109) XI
C..... TEST IF PRINTOUT TO BE PRODUCED
C
IF((NITER+NPRINT-IP)/NPRINT.NE.NITER/NPRINT) GO TO 10
WRITE(6,103) (ASYMBL(K),K=1,7)
WRITE(6,104) NITER, (RSDU(K),K=1,7), NWJ, NWI, NFJ, NFI, NTJ, NTI
WRITE(6,107) (ASYMBL(K),K=8,14)
WRITE(6,108) NITER, (SAN(K),K=1,7), NKJ, NKI, NEJ, NEI
C
WRITE(15,108) NITER, (SAN(K),K=1,7), NKJ, NKI, NEJ, NEI
WRITE(6,109) XI
C
WRITE(15,109) XI
10 CONTINUE
C..... TEST IF MAXIMUM NUMBER OF ITERATIONS PERFORMED
C
IF(NITER.EQ.NMAX) GO TO 8
RES=0.
DO 7 K=1, IE
IF(ABS(RES).LT.ABS(RSDU(K))) RES=RSDU(K)
RSDU(K)=0.
7 CONTINUE
SANWF=0.
DO 3 K=1, IE
IF(ABS(SANWF).LT.ABS(SAN(K))) SANWF=SAN(K)
SAN(K)=0.
3 CONTINUE
C*** TEST IF CONVERGENCE CRITERION SATISFIED; IF NOT, BACK TO(1)
IF(ABS(SANWF).GT..001) GO TO 1
IF (NITER.LE.NMIN) GO TO 1
C..... "CONVERGED"
C
WRITE(6,105) NITER
GO TO 340
C..... "UNCONVERGED"
C
8 WRITE(6,106) NITER
FOLLOW UP CALCULATIONS
C
**-----**
340 CONTINUE
CALL CONLOT(XI,PERFIN,PCASC)
C
CALCULATE THE LOCAL MIXING TIMES
C
-----
DEPTH=X(5)-X(1)

```

```

EMFO1360
EMFO1370
EMFO1380
EMFO1390
EMFO1400
EMFO1410
EMFO1420
EMFO1430
EMFO1440
EMFO1450
EMFO1460
EMFO1470
EMFO1480
EMFO1490
EMFO1500
EMFO1510
EMFO1520
EMFO1530
EMFO1540
EMFO1550
EMFO1560
EMFO1570
EMFO1580
EMFO1590
EMFO1600
EMFO1610
EMFO1620
EMFO1630
EMFO1640
EMFO1650
EMFO1660
EMFO1670
EMFO1680
EMFO1690
EMFO1700
EMFO1710
EMFO1720
EMFO1730
EMFO1740
EMFO1750
EMFO1760
EMFO1770
EMFO1780
EMFO1790
EMFO1800

```

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```

          DIMNSN=DEPTH
          IL=I1+1
          IH=I3-1
          DO 900 J=1,JNM
          DO 900 I=IL,IH
          TIMIX(I,J) = 80.0*(DIMNSN**2/(1000.*A(I,J,NEP)))*0.333
900      CONTINUE
C      CALCULATE THE TOTAL MIXING TIME FOR VESSEL
C      -----
          XMASS=RM**2*DEPTH*RDFREF(1)/2.0
          PSPEC=PCASC/XMASS
          TOTALT=80.0*(DIMNSN**2/(ETA*1000.*PSPEC))*0.333
C      -----
C      CALCULATE THE POWER INPUT PER TON MELT
C      -----
          PTON = PSPEC* 1000.0
C      -----
C.....MAIN RESULTS PRINTED OUT HERE
C.....
          IF(PERFIN.GT.PIMIN)WRITE(6,250) PERFIN
          IF(PERFIN.LE.PIMIN) WRITE(6,252) PERFIN
          WRITE(6,254) PTON
          WRITE(6,253) TOTALT
          CALL PRINT(NW,NF,2)
          CALL PRINT(NV1,NV2,1)
          IF (NTURB.EQ.0) GO TO 355
C.....TURBULENT RESULTS PRINT OUT
          CALL PRINT(NK,NEP,2)
          CALL PRINT(NMU,NMU,2)
C      -----
C      WRITE OUTPUT DATA FILE
C      -----
355      CONTINUE
          DO 15 K=NW,IE
          DO 16 J=1,JN
16      WRITE(7,14)(A(I,J,K),I=I1,I3)
15      CONTINUE
C
          DO 11 K=NHR,LV
          DO 11 J=1,JN
          WRITE(3,14)(A(I,J,K),I=1,IN)
11      CONTINUE
C      -----
C      CALCULATE GENERATION AND DISSIPATION ENERGY
C      -----

```

```

EMFO1810
EMFO1820
EMFO1830
EMFO1840
EMFO1850
EMFO1860
EMFO1870
EMFO1880
EMFO1890
EMFO1900
EMFO1910
EMFO1920
EMFO1930
EMFO1940
EMFO1950
EMFO1960
EMFO1970
EMFO1980
EMFO1990
EMFO2000
EMFO2010
EMFO2020
EMFO2030
EMFO2040
EMFO2050
EMFO2060
EMFO2070
EMFO2080
EMFO2090
EMFO2100
EMFO2110
EMFO2120
EMFO2130
EMFO2140
EMFO2150
EMFO2160
EMFO2170
EMFO2180
EMFO2190
EMFO2200
EMFO2210
EMFO2220
EMFO2230
EMFO2240
EMFO2250

```

253

```

C...ENERGY GENERATION:
C -----
50      CALL VOL(GK,PGEN)
C...DISSIPATION:
C -----
          DO 52 I=1,IN
          DO 52 J=1,JN
          EPSIL(I,J)=0.0
52      CONTINUE
          IL=I1+1
          IH=I3-1
          DO 400 I=IL,IH
          DO 400 J=2,JNM
          EPSIL(I,J)=A(I,J,NEP)
400     CONTINUE
          CALL VOL(EPSIL,PDISIP)
          PDISIP=PDISIP*ROREF(1)
C-----
C...PRINT IT:
C -----
          WRITE(6,410) PGEN,PDISIP,PCASC
410     FORMAT('TOTAL GENERATION= ',1PE10.2/'TOTAL DISSIPATION= '
          1,1PE10.2/'TOTAL POWER INPUT= ',1PE10.2)
C-----
C***** S T O P *****
969    CONTINUE
C *** SAVE DATA FOR PLOTTING (UNIT #9)
      NX=I3-I1+1
      NY=JN
      WRITE(9,1233) NX,NY,J1,NSCHEM,CU,ETA
C          COORDINATE DATA
C -----
      DO 34 I=I1,I3
      L=I-I1+1
34     XM(L)=X1(I)-X1(I1)
      WRITE(9,1234) (XM(L),L=1,NX)
      WRITE(9,1234) (X2(J),J=1,JN)
          K=NUV1
C          -----
      DO 32 J=1,NY
32     WRITE(9,1234) (A(I,J,K),I=I1,I3)
          K=NUV2
C          -----
      DO 33 J=1,NY
33     WRITE(9,1234)(A(I,J,K),I=I1,I3)

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```

EMFO2260
EMFO2270
EMFO2280
EMFO2290
EMFO2300
EMFO2310
EMFO2320
EMFO2330
EMFO2340
EMFO2350
EMFO2360
EMFO2370
EMFO2380
EMFO2390
EMFO2400
EMFO2410
EMFO2420
EMFO2430
EMFO2440
EMFO2450
EMFO2460
EMFO2470
EMFO2480
EMFO2490
EMFO2500
EMFO2510
EMFO2520
EMFO2530
EMFO2540
EMFO2550
EMFO2560
EMFO2570
EMFO2580
EMFO2590
EMFO2600
EMFO2610
EMFO2620
EMFO2630
EMFO2640
EMFO2650
EMFO2660
EMFO2670
EMFO2680
EMFO2690
EMFO2700

```

254

```

30 CONTINUE
      K=NK
C
DO 121 J=1,JN
WRITE(9,1234) (A(I,J,K),I=I1,I3)
121 CONTINUE
      K=NEP
C
DO 122 J=1,JN
122 WRITE(9,1234)(A(I,J,K),I=I1,I3)
      K=NMU
C
DO 123 J=1,JN
123 WRITE(9,1234)(A(I,J,K),I=I1,I3)
      LOCAL MIXING TIMES
C
DO 124 J=1,JN
124 WRITE(9,1234)(TIMIX(I,J),I=I1,I3)
STOP
C-----
C          F O R M A T T I N G
C-----
9  FORMAT(////'          C O O R D I N A T E      D A T A '//)
12 FORMAT(4H I1=,I2,4H I2=,I2,4H I3=,I2,4H IA=,I2,4H IN=,I2
1,4H J1=,I2,4H JN=,I2)
14 FORMAT(7(1PE10.3))
40 FORMAT(27HOTHE ELECTRODE TIP SHAPE IS/
125X,5HX2(J),25X,6HX1(II))
42 FORMAT(20X,E15.5,15X,E15.5)
101 FORMAT(25HODISTANCES IN DIRECTION-1/(1H ,7E15.5))
102 FORMAT(25HODISTANCES IN DIRECTION-2/(1H ,7E15.5))
103 FORMAT(36HOMAXIMUM RESIDUAL FOR EACH VARIABLE://
16HONITER,7(3X,A6),7X,'NWJ NWI NFJ NFI NTJ NTI'//)
104 FORMAT(1H0,I3,3X,7(F9.4),5X,I2,5(I4))
105 FORMAT(32HOTHE PROCESS          CONVERGED IN,I5,13H  ITERATIONS)
106 FORMAT(32HOTHE PROCESS DID NOT CONVERGE IN,I5,13H  ITERATIONS)
107 FORMAT(36HOMAXIMUM RESIDUAL FOR EACH VARIABLE://
16HONITER,7(3X,A6),7X,'NKJ NKI NEJ NEI'//)
108 FORMAT(1H0,I3,3X,7(F9.4),5X,I2,3(I4))
109 FORMAT(' XI= ',1PE10.2)
200 FORMAT( 6A6)
205 FORMAT(5I5,F5.2,6I5)
208 FORMAT('          I N P U T      D A T A '//)
210  FORMAT('CURRENT= ',1PE10.2,' KILO-AMPS'
1/'MOLECULAR VISCOSITY= ',1PE10.2,' KG/M-S'
EMFO2710
EMFO2720
EMFO2730
EMFO2740
EMFO2750
EMFO2760
EMFO2770
EMFO2780
EMFO2790
EMFO2800
EMFO2810
EMFO2820
EMFO2830
EMFO2840
EMFO2850
EMFO2860
EMFO2870
EMFO2880
EMFO2890
EMFO2900
EMFO2910
EMFO2920
EMFO2930
EMFO2940
EMFO2950
EMFO2960
EMFO2970
EMFO2980
EMFO2990
EMFO3000
EMFO3010
EMFO3020
EMFO3030
EMFO3040
EMFO3050
EMFO3060
EMFO3070
EMFO3080
EMFO3090
EMFO3100
EMFO3110
EMFO3120
EMFO3130
EMFO3140
EMFO3150

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2//MELT DENSITY= ',1PE10.2,' KG/(M-CUBE)'
3//ELEC. CONDUCTIVITY OF MELT=',1PE10.2,' MHO-M'
4//MAX.ITERATIONS FOR FLOW ',I5)
213   FORMAT('COLD START CALCULATION; DATA NOT READ IN')
214   FORMAT('CONTINUATION CALCULATION;PREVIOUS DATA READ')
215   FORMAT('CURRENT TYPE IS ==>A.C.')
216   FORMAT('CURRENT TYPE IS ==>D.C.')
217   FORMAT('L A M I N A R   FLOW CALCULATION')
218   FORMAT('T U R B U L E N T   FLOW CALCULATION')
229   FORMAT('K-EPSILON TURBULENCE MODEL')
230   FORMAT('SPECRAL MODEL FOR TURBULENCE')
231   FORMAT('E-M FIELD CALCULATIONS => O F F ')
232   FORMAT('E-M FIELD CALCULATIONS => O N ')
233   FORMAT('FLOW CALCULATIONS      => O F F ')
234   FORMAT('FLOW CALCULATIONS      => O N ')
235   FORMAT('USING LOW REYNOLD NUMBER MODEL ')
236   FORMAT('BASE VALUE OF CONSTANTS ARE:/'
1'CD=      ',F5.2/
2'C1=      ',F5.2/
3'C2=      ',F5.2)
250   FORMAT('//' AUTOMATIC TURBULENCE CHECK; FLOW FOUND TO BE:'
1/10X,'.....T U R B U L E N T '
2/10X,'PERFORMANCE INDEX= ',F7.2)
252   FORMAT(' AUTOMATIC TURBULENCE CHECK; FLOW FOUND TO BE:'
1/10X,'.....L A M I N A R '
2/10X,'PERFORMANCE INDEX= ',F7.2)
254   FORMAT(10X,'POWER INPUT= ',1PE10.2,' WATTS/TON OF MELT')
253   FORMAT(10X,'OVERALL MIXING TIME(95% MIXED)= ',F6.2,' SECS'//)
220   FORMAT('ELECTRODE RADIUS= ',1PE10.2,' M'
1//VESSEL RADIUS= ',1PE10.2,' M'
2//MELT DEPTH= ',1PE10.2,' M'
3//ELECTRODE IMMERSION= ',1PE10.2,' M')
225   FORMAT(/,50H DRIVING FORCES ARE FIRST ELECTROMAG THEN BUOYANCY)
1233  FORMAT(4I4,2F5.2)
1234  FORMAT(5E13.5)
      END

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```

EMFO3160
EMFO3170
EMFO3180
EMFO3190
EMFO3200
EMFO3210
EMFO3220
EMFO3230
EMFO3240
EMFO3250
EMFO3260
EMFO3270
EMFO3280
EMFO3290
EMFO3300
EMFO3310
EMFO3320
EMFO3330
EMFO3340
EMFO3350
EMFO3360
EMFO3370
EMFO3380
EMFO3390
EMFO3400
EMFO3410
EMFO3420
EMFO3430
EMFO3440
EMFO3450
EMFO3460
EMFO3470
EMFO3480
EMFO3490
EMFO3500
EMFO3510

```





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```

1 CONTINUE
C
C      UPDATE COEFFICIENT CD TO CDMOD(I,J) FOR
C      LOW REYNOLDS NUMBER MODEL OF TURBULENCE.
C      -----
          IF(NTURB.EQ.0) GO TO 500
          DO 500 J=1,JNM
            IL=I1+1
            IH=I3-1
            DO 500 I=IL,IH
              FMU=1.0
              IF(LREYN.EQ.0.OR.A(I,J,NEP).EQ.0.0) GO TO 400
              REYLOC=A(I,J,NRO)*A(I,J,NK)**2/(ZMUREF(1)*A(I,J,NEP))
              FMU=EXP(-2.5/(1+REYLOC/50.0))
400          CONTINUE
          CDMOD(I,J)=CD*FMU
500          CONTINUE
C-----
C.....OBTAIN EFFECTIVE VISCOSITY
C-----
          IF (NTURB.EQ.1) CALL VISCOS
C.....VORTICITY SUB-CYCLE
C-----
          GOSA=0.0
          TOTA=0.0
          CALL VORITY
          IF(GOSA.EQ.0.0.AND.TOTA.EQ.0.0) GO TO 10
          SAN(NW)=GOSA/TOTA
          GO TO 40
10 SAN(NW)=1.
40 CONTINUE
C.....STREAM FUNCTION SUB-CYCLE
C-----
          GOSA=0.0
          TOTA=0.0
          CALL STRFUN
          IF(GOSA.EQ.0.0.AND.TOTA.EQ.0.0) GO TO 20
          SAN(NF)=GOSA/TOTA
          GO TO 50
20 SAN(NF)=1.
50 CONTINUE
C.....VELOCITY DISTRIBUTION
C-----
          CALL VELDIS
          IF(NTURB.EQ.0) GO TO 120
C.....TURBULENT VARIABLE CALCULATIONS(ONLY IF NTURB=1)

```

```

EQN00460
EQN00470
EQN00480
EQN00490
EQN00500
EQN00510
EQN00520
EQN00530
EQN00540
EQN00550
EQN00560
EQN00570
EQN00580
EQN00590
EQN00600
EQN00610
EQN00620
EQN00630
EQN00640
EQN00650
EQN00660
EQN00670
EQN00680
EQN00690
EQN00700
EQN00710
EQN00720
EQN00730
EQN00740
EQN00750
EQN00760
EQN00770
EQN00780
EQN00790
EQN00800
EQN00810
EQN00820
EQN00830
EQN00840
EQN00850
EQN00860
EQN00870
EQN00880
EQN00890
EQN00900

```

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C ----- EQNO0910
C CHOOSE WHICH SCHEME OF CALCULATION IS DESIRED. EQNO0920
C      NSCHEM=0 ...FOR K-EPSILON MODEL EQNO0930
C      NSCHEM=1 ...FOR MSR MODEL. EQNO0940
C ----- EQNO0950
C      CALL WALL EQNO0960
C      NEND=NK EQNO0970
C      IF(NSCHEM.EQ.0) NEND=NEP EQNO0980
C EQNO0990
C      DO 41 K=NK,NEND EQNO1000
C      GOSA=0.0 EQNO1010
C      TOTA=0.0 EQNO1020
C      IF(NSCHEM.EQ.0) CALL TURVAR(K) EQNO1030
C      IF(NSCHEM.EQ.1) CALL TURVAR(K) EQNO1040
C      IF(GOSA.EQ.0.0.AND.TOTA.EQ.0.0) GO TO 30 EQNO1050
C      SAN(K)=GOSA/TOTA EQNO1060
C      GO TO 41 EQNO1070
C      30 SAN(K)=1. EQNO1080
C      41 CONTINUE EQNO1090
C..... INITIATE ITERATION ON BOUNDARY NODES EQNO1100
C ----- EQNO1110
C 120 CALL BOUND EQNO1120
C..... CALL DISSIP IF SO DESIRED BY CHOICE OF SCHEME. EQNO1130
C ----- EQNO1140
C      IF (NTURB.EQ.1.AND.NSCHEM.EQ.1)CALL DISSIP EQNO1150
C      OUTPUT CDMOD(I,J) AS A CHECK EQNO1160
C ----- EQNO1170
C      DO 550 J=1,JNM EQNO1180
C      IL=I+1 EQNO1190
C      IH=I3-1 EQNO1200
C      WRITE(18,560)(CDMOD(I,J),I=IL,IH) EQNO1210
550 CONTINUE EQNO1220
560 FORMAT(10F5.2) EQNO1230
C RETURN EQNO1240
C END EQNO1250

```



```

1 CONTINUE
GOSA=0.
TOTA=0.
C CONVECTION HAS ZERO EFFECT ON MAGNETIC DIFFUSION EQN.
      AE=0.
      AW=0.
      AN=0.
      AS=0.
C-----
C CALCULATE MAGNETIC INDUCTION AT INTERIOR POINTS
C-----
LV=NHI
IF(LTYP.EQ.2) LV=NHR
      DO 17 K=NHR,LV
      DO 11 J=2,JNM

IL=IMIN(J)
IF(J.EQ.J1) IL=I1+1
IH=IMAX(J)
C BRANCH TO POINTS ON ELECTRODE FACES
C-----
      DO 19 I=IL,IH
      IF(J.EQ.J1.AND.I.LE.I2) GO TO 30
      IF(I.EQ.IW(J).AND.J.LT.J1) GO TO 15
      IF(I.EQ.I3) GO TO 14

CALL SORCE(SOURCE,I,J,K)
RISQ=1./R(J)/R(J)
SOP=A(I,J,NS)
BBE=4./(A(I+1,J,NS)+SOP)*RISQ*BE(I)
IF(I.EQ.(IW(J)-1).AND.J.LT.J1) BBE=4./(SOP+SOP)*RISQ*BE(I)
BBW=4./(A(I-1,J,NS)+SOP)*RISQ*BW(I)
IF(I.EQ.(I3+1)) BBW=4./(SOP+SOP)*RISQ*BW(I)
BBN=16./(A(I,J+1,NS)+SOP)/((R(J+1)+R(J))**2)*BN(J)
IF(J.EQ.(J1-1).AND.I.LT.I2) BBN=BBN*(A(I,J+1,NS)+SOP)/(2.*SOP)
IF(I.EQ.(IW(J)-1).AND.J.LT.J1) BBN=BBN*(A(I,J+1,NS)+SOP
1)/(2.*SOP)
BBS=16./(A(I,J-1,NS)+SOP)/((R(J-1)+R(J))**2)*BS(J)
ANUM=(AE+R(J)*R(J)*BBE)*A(I+1,J,K)
1+(AW+R(J)*R(J)*BBW)*A(I-1,J,K)
2+(AN+R(J+1)*R(J+1)*BBN)*A(I,J+1,K)
3+(AS+R(J-1)*R(J-1)*BBS)*A(I,J-1,K)+SOURCE
ADNM=AE+AW+AN+AS+R(J)*R(J)*(BBE+BBW+BBN+BBS)
GO TO 21
C-----
C POINTS ON ELECTRODE FACES
C-----

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FIE00460
FIE00470
FIE00480
FIE00490
FIE00500
FIE00510
FIE00520
FIE00530
FIE00540
FIE00550
FIE00560
FIE00570
FIE00580
FIE00590
FIE00600
FIE00610
FIE00620
FIE00630
FIE00640
FIE00650
FIE00660
FIE00670
FIE00680
FIE00690
FIE00700
FIE00710
FIE00720
FIE00730
FIE00740
FIE00750
FIE00760
FIE00770
FIE00780
FIE00790
FIE00800
FIE00810
FIE00820
FIE00830
FIE00840
FIE00850
FIE00860
FIE00870
FIE00880
FIE00890
FIE00900

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GOSA=GOSA+ABS(Z-A(I,J,K)) FIE01360
TOTA=TOTA+ABS(A(I,J,K)) FIE01370
C .....STORE MAXIMUM RESIDUAL FIE01380
C ..... FIE01390
IF(ABS(RS).LE.ABS(RSDU(K))) GO TO 6 FIE01400
RSDU(K)=RS FIE01410
6 CONTINUE FIE01420
19 CONTINUE FIE01430
11 CONTINUE FIE01440
IF(GOSA.EQ.0.0.AND.TOTA.EQ.0.0) GO TO 7 FIE01450
SAN(K)=GOSA/TOTA FIE01460
GO TO 8 FIE01470
7 SAN(K)=1. FIE01480
8 CONTINUE FIE01490
17 CONTINUE FIE01500
C ----- FIE01510
C BOUNDARY CONDITIONS(I=1,J<J1) FIE01520
C ----- FIE01530
J11=J1-1 FIE01540
XQ=X1(2)-X1(1) FIE01550
XR=X1(3)-X1(1) FIE01560
BB=XR*XR/(XR*XR-XQ*XQ) FIE01570
DO 10 K=NHR,NHI FIE01580
DO 10 J=2,J11 FIE01590
10 A(1,J,K)=BB*A(2,J,K)+(1.-BB)*A(3,J,K) FIE01600
C .....BOUNDARY I=IN FIE01610
C ----- FIE01620
XQ=X1(IN)-X1(INM) FIE01630
XR=X1(IN)-X1(INM-1) FIE01640
BB=XR*XR/(XR*XR-XQ*XQ) FIE01650
DO 151 K=NHR,NHI FIE01660
DO 151 J=2,JNM FIE01670
151 A(IN,J,K)=BB*A(INM,J,K)+(1.-BB)*A(INM-1,J,K) FIE01680
CONTINUE FIE01690
C ..... J=1, 1<I<IN FIE01700
C ----- FIE01710
DO 25 K=NHR,NHI FIE01720
DO 25 I=1,IN FIE01730
25 A(I,1,K)=A(I,2,K) FIE01740
CONTINUE FIE01750
IF(LI.EQ.2) GO TO 9 FIE01760
C ----- FIE01770
C CONVERGENCE CRITERION CHECK FIE01780
C ----- FIE01790
CF1=ABS(RSDU(NHR)) FIE01800

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CF2=ABS(RSDU(NHI))
CFF=CF1
IF(CF1.LT.CF2) CFF=CF2
IF(CFF.LE.CP) GO TO 9
NIT=NIT+1
RSDU(NHR)=0.
RSDU(NHI)=0.
-----
C          PRINTOUT CHECK
C          -----
C          K1=NIT/MPRINT
          K2=K1*MPRINT
          IF(NIT.NE.K2) GO TO 5
          WRITE(6,104) NIT,CF1,CF2
104  FORMAT(/,6H NIT=,I4/,6H CF1=,E14.8,6H CF2=,E14.8/)
          5          CONTINUE
-----
C          MAXIMUM ITERATION CHECK
C          -----
C          IF(NIT.LT.MIT) GO TO 1
          WRITE(6,106) NIT
106  FORMAT(32H0THE FLD.EQN DID NOT CONVERGE IN,15,13H ITERATIONS)
          LF=2
          GO TO 20
-----
C          CALCULATE CURRENT DENSITY
C          -----
          9          CONTINUE
          DO 29 J=1,JN
          DO 191 I=1,IN
          IF(I.LT.I1.AND.J.GT.J1) GO TO 191
          IF(I.EQ.1.OR.I.EQ.IN) GO TO 192
          IF(J.EQ.J1.AND.I.LE.I1) GO TO 192
          IF(J.EQ.1.OR.J.EQ.JN) GO TO 192
          A(I,J,NJRR)=-ADF(I,J,1,NHR)*R(J)
          A(I,J,NJRI)=-ADF(I,J,1,NHI)*R(J)
          IF(I.EQ.I1.AND.J.GT.J1) GO TO 191
192          A(I,J,NJZR)=R(J)*ADF(I,J,2,NHR)+A(I,J,NHR)*2.
          191          A(I,J,NJZI)=R(J)*ADF(I,J,2,NHI)+A(I,J,NHI)*2.
          191          CONTINUE
          29          CONTINUE
-----
C          MODIFICATIONS FOR CURRENT AT THE BOUNDARIES
C          -----
C          AT I=IW ,J<J1

```

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FIEO1810
FIEO1820
FIEO1830
FIEO1840
FIEO1850
FIEO1860
FIEO1870
FIEO1880
FIEO1890
FIEO1900
FIEO1910
FIEO1920
FIEO1930
FIEO1940
FIEO1950
FIEO1960
FIEO1970
FIEO1980
FIEO1990
FIEO2000
FIEO2010
FIEO2020
FIEO2030
FIEO2040
FIEO2050
FIEO2060
FIEO2070
FIEO2080
FIEO2090
FIEO2100
FIEO2110
FIEO2120
FIEO2130
FIEO2140
FIEO2150
FIEO2160
FIEO2170
FIEO2180
FIEO2190
FIEO2200
FIEO2210
FIEO2220
FIEO2230
FIEO2240
FIEO2250

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C          -----
C          DO 27 J=2,J1
C          II=IW(J)
C          A(II,J,NJRR)=(A(II,J,NHR)-A(II+1,J,NHR))/(X1(II+1)-X1(II))*R(J)
C          A(II,J,NJRI)=(A(II,J,NHI)-A(II+1,J,NHI))/(X1(II+1)-X1(II))*R(J)
C          A(II,J,NJZR)=(R(J+1)*R(J+1)*A(II,J+1,NHR)-R(J)*R(J)*A(II,J
1,NHR))/(X2(J+1)-X2(J))/R(J)
C          A(II,J,NJZI)=(R(J+1)*R(J+1)*A(II,J+1,NHI)-R(J)*R(J)*A(II
1,J,NHI))/(X2(J+1)-X2(J))/R(J)
27          CONTINUE
C          .....AT I=I3
C          -----
C          DO 28 J=2,JNM
C          A(I3,J,NJRR)=(A(I3-1,J,NHR)-A(I3,J,NHR))/(X1(I3)-X1(I3-1))*R(J)
C          A(I3,J,NJRI)=(A(I3-1,J,NHI)-A(I3,J,NHI))/(X1(I3)-X1(I3-1))*R(J)
28          CONTINUE
C          ..... AT J=J1 ,I1<I<I2
C          -----
C          IF(I1.EQ.I2) GO TO 32
C          DO 31 I=I1,I2
C          A(I,J1,NJZR)=(R(J1+1)*R(J1+1)*A(I,J1+1,NHR)-R(J1)*R(J1)*A(I,J1,
1NHR))/(X2(J1+1)-X2(J1))/R(J1)
C          A(I,J1,NJZI)=(R(J1+1)*R(J1+1)*A(I,J1+1,NHI)-R(J1)*R(J1)*A(I,J1
1,NHI))/(X2(J1+1)-X2(J1))/R(J1)
31          CONTINUE
32          CONTINUE
C          DO 115 I=1,IN
C          DO 115 J=1,JN
C          IF(J.GT.J1.AND.I.LT.I1) GO TO 115
C          A(I,J,NJJ)=(A(I,J,NJRR)**2+A(I,J,NJRI)**2+A(I,J,NJZR)**2+A(I,J,
1NJZI)**2)
C          A(I,J,NJJ)=500.*A(I,J,NJJ)/A(I,J,NS)
115          CONTINUE
C          .....CALCULATE VOLTAGE
C          -----
C          DO 200 I=I1,I3
C          N=I-I1+1
C          JL=J1
C          IF(I.GT.I2.AND.I.LE.IW(1)) JL=J1-(I-I2)
C          IF(I.GT.IW(1)) JL=1
C          DO 201 J=1,JN
201          AVM(J)=R(J)*A(I,J,NJJ)
200          FUN(N)=FINT(JL,JN,2,1,1)
C          N3=I3-I1+1
C          DO 202 N=1,N3

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FIE02260
FIE02270
FIE02280
FIE02290
FIE02300
FIE02310
FIE02320
FIE02330
FIE02340
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FIE02370
FIE02380
FIE02390
FIE02400
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FIE02470
FIE02480
FIE02490
FIE02500
FIE02510
FIE02520
FIE02530
FIE02540
FIE02550
FIE02560
FIE02570
FIE02580
FIE02590
FIE02600
FIE02610
FIE02620
FIE02630
FIE02640
FIE02650
FIE02660
FIE02670
FIE02680
FIE02690
FIE02700

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202          AVM(N)=FUN(N)          FIE02710
SUMI=FINT(1,N3,1,I1,1)          FIE02720
POWR=2.*PI*SUMI                FIE02730
VOLT=POWR*1.414/CU             FIE02740
CURR=CU/1.414                  FIE02750
          IF(LI.EQ.2) GO TO 37    FIE02760
WRITE(6,110) NIT                FIE02770
CALL PRINT(NHR,LV,2)           FIE02780
37          CONTINUE              FIE02790
WRITE(6,204) VOLT,CURR          FIE02800
110 FORMAT(32H THE FIELD EQUATION CONVERGED IN,15,13H ITERATIONS) FIE02810
204 FORMAT(/,9H VOLTAGE=,F10.3,9H CURRENT=,F10.4) FIE02820
20 CONTINUE                      FIE02830
RETURN                          FIE02840
END                              FIE02850
```

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```
FUNCTION FINT(NL,NU,ND,NI,NJ)          FIN00010
COMMON/CCORD/IMIN(21),IMAX(21),X1(31),X2(21),R(21)  FIN00020
COMMON/CGRID/IN,INM,JN,JNM,I1,I2,I3,IA,IW(21),J1    FIN00030
COMMON/CAVM/AVM(60)                             FIN00040
NU1=NU-1                                          FIN00050
SUMA=0.                                           FIN00060
DO 10 N=NL,NU1                                   FIN00070
IF(ND.EQ.2) GO TO 1                              FIN00080
I=N+NI-1                                         FIN00090
DX=0.5*(X1(I+1)-X1(I))                          FIN00100
GO TO 2                                          FIN00110
1 J=N+NJ-1                                       FIN00120
DX=0.5*(X2(J+1)-X2(J))                          FIN00130
2 CONTINUE                                       FIN00140
10 SUMA=SUMA+(AVM(N)+AVM(N+1))*DX               FIN00150
FINT=SUMA                                        FIN00160
RETURN                                           FIN00170
END                                              FIN00180
```



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C	-----	INI00460
	DO 30 K=1,20	INI00470
	DO 30 J=1,JN	INI00480
	DO 30 I=1,IN	INI00490
	A(I,J,K)=0.0	INI00500
30	CONTINUE	INI00510
C	-----	INI00520
C	MAGNETIC INDUCTION GUESS FIELD	INI00530
C	-----	INI00540
	DO 31 J=1,J1	INI00550
	DO 31 I=1,I2	INI00560
31	A(I,J,NHR)=CU/(2.0*PI*RE*RE)	INI00570
C	-----	INI00580
C	GUESS FIELD FOR TEMPERATURE	INI00590
C	-----	INI00600
	DO 50 J=1,JN	INI00610
	DO 50 I=1,I3	INI00620
	IF(J.GT.J1.AND.I.LT.I1) GO TO 50	INI00630
	A(I,J,NT)=300.0	INI00640
	IF(J.LE.J1.AND.I.LE.I2) A(I,J,NT)=300.0	INI00650
	IF(I.EQ.I3) A(I,J,NT)=300.0	INI00660
50	CONTINUE	INI00670
C	-----	INI00680
C	FIXED BOUNDARY CONDITIONS	INI00690
C	-----	INI00700
	DO 11 I=I1,IN	INI00710
11	A(I,JN,NHR)=CU/(2.0*PI*RM*RM)	INI00720
	J12=J1+1	INI00730
C		INI00740
	DO 12 J=J12,JN	INI00750
	A(I1,J,NHR)=CU/(2.0*PI*R(J)*R(J))	INI00760
12	CONTINUE	INI00770
C		INI00780
	DO 20 J=2,JNM	INI00790
	IL=I1+1	INI00800
	IF(J.LE.J1) IL=IW(J)+1	INI00810
	IH=I3-1	INI00820
	DO 20 I=IL,IH	INI00830
	A(I,J,NK)=1.0E-05	INI00840
	A(I,J,NEP)=1.0E-06	INI00850
20	CONTINUE	INI00860
C		INI00870
	DO 14 J=1,JN	INI00880
	DO 14 I=1,I3	INI00890
14	SVOR(I,J)=0.	INI00900

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C
17 DO 17 I=I1,IH
RETURN
END
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INIO0910
INIO0920
INIO0930
INIO0940
INIO0950
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SUBROUTINE PRINT(NBEGIN,NTOTAL,LP)	PRI00010
REAL*8 ANAME(6,20),ASYMBL(14)	PRI00020
\$	PRI00030
COMMON/CNUMB/NW,NF,NK,NEP,NV1,NV2,NT,NMU,NRO,NTC,NSP,NMT	PRI00040
COMMON/CCORD/IMIN(21),IMAX(21),X1(31),X2(21),R(21)	PRI00050
COMMON/CGRID/IN,INM,JN,UNM,I1,I2,I3,IA,IW(21),J1	PRI00060
COMMON/CDIM/X(5),RE,RM	PRI00070
COMMON/CITER/NMAX,NPRINT,NITER,IP,IE,IV,CC	PRI00080
COMMON/CCONT/C1,C2,C3,CD	PRI00090
COMMON/CRLAX/RP(10),RSDU(10),SAN(10)	PRI00100
COMMON/CEMP/WF,P,S(4)	PRI00110
COMMON/CHJ/NHR,NHI,NJRR,NJRI,NJZR,NJZI,NJJ,NS	PRI00120
COMMON/CTHERM/TCREF(5),SPREF(5),BETA(2),ES,EE,EM,SB,HL	PRI00130
COMMON/CTEMP/TLM,TSM,TLS,TA,TM,TW	PRI00140
COMMON/CTRVL/VE,VC	PRI00150
COMMON/CHEAT/HC,HW(4)	PRI00160
COMMON/CPROP/ROREF(5),ZMUREF(2),PR(10),GAMA	PRI00170
COMMON/CVF/FES(21),FEM(21),FSM,FSE	PRI00180
COMMON/CDROP/TAU,HCD,D,QS,QM	PRI00190
COMMON/CTAUW/TAUW1(20),TAUW2(15),TAUW3(21),TAUW4(10),TAUW5(21)	PRI00200
1,CAPPA,E	PRI00210
COMMON/CFLD/CU	PRI00220
COMMON/CPI/PI	PRI00230
COMMON/CTYP/LTYP	PRI00240
COMMON/CANG/THETA	PRI00250
COMMON/FITER/MIT,MPRINT,CP	PRI00260
COMMON/CDVAR/A(31,21,20)	PRI00270
COMMON/CNAME/ANAME,ASYMBL	PRI00280
COMMON/CCRIT/NWI,NWJ,NFI,NFJ,NTI,NTJ,NKI,NKJ,NEI,NEJ,GOSA,TOTA	PRI00290
COMMON/CLF/LF,LI	PRI00300
COMMON/CSVOR/SVOR(31,21)	PRI00310
COMMON/CTURB/NTURB,NSCHEM,NVISC,RPXI,XI	PRI00320
COMMON/CSNW/SN1(20),SN2(15),SN3(21),SN4(20),SN5(21)	PRI00330
COMMON/CB/BE(31),BW(31),BN(21),BS(21)	PRI00340
COMMON/CAVM/AVM(60)	PRI00350
COMMON/CHBL/HLW,HLEH,HLEV,HLS,HLD,HLM,POWR	PRI00360
COMMON/CSUR/HS(2),TS(2)	PRI00370
COMMON/CBB/BBE,BBW,BBN,BBS,BPP	PRI00380
COMMON/CSORSE/SOURCE,SPRIME	PRI00390
COMMON/CGEOM/FR1,FR2(15),FR3,FR4,FR5	PRI00400
COMMON/CAF/FUN(20)	PRI00410
COMMON/CVOLT/VOLT	PRI00420
COMMON/CAVT/TB	PRI00430
COMMON/CYP/YP1(20),YP2(15),YP3(21),YP4(10),YP5(21)	PRI00440
\$	PRI00450

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JX=JN/10
IF(JX.LT.1) JX=1
IX=IN/10
IF(IX.LT.1) IX=1
DO 10 K=NBEGIN,NTOTAL
WRITE(6,100) (ANAME(L,K),L=1,6)
IB=I3
JX=1
DO 2 L=1,JN,JX
J=JN+1-L
WRITE(6,103) (A(I,J,K),I=I1,IB,1)
WRITE(6,105) J
2 CONTINUE
WRITE(6,104) (I,I=I1,IB,1)
10 CONTINUE
IF(LP.EQ.2) GO TO 20
C WRITE(6,1) HLW,HLEH,HLEV,HLS,HLD,HLM,POWR,VOLT
N1=I3-I1-1
J12=J1+1
C WRITE(6,6) (SN1(N),N=1,N1)
C WRITE(6,7) (SN2(N),N=2,J1)
C WRITE(6,8) (SN3(N),N=2,JNM)
WRITE(6,15) (YP1(N),N=1,N1)
WRITE(6,16) (YP2(N),N=2,J1)
WRITE(6,17) (YP3(N),N=2,JNM)
WRITE(6,19) (YP5(J),J=J12,JNM)
C WRITE(6,212) VE,TB
20 CONTINUE
RETURN
100 FORMAT(1H1,30X,21HTHE DISTRIBUTION OF ,6A6/
126X,51H-----/
21HO,115X,1HJ/115X,3H---//)
1 FORMAT(/,5H HLW=,E10.3,6H HLEH=,E10.3,6H HLEV=,E10.3
1,5H HLS=,E10.3,5H HLD=,E10.3,5H HLM=,E10.3,6H POWR=,E10.3
2/6H VOLT=,F10.3)
103 FORMAT(/(1H ,10(1PE10.3)))
104 FORMAT(4H I/(1H ,7X,9(I2,8X),I2))
105 FORMAT(105X,I2)
201 FORMAT(/,32H TEMP. DISTRIBUTION IN ELECTRODE)
203 FORMAT(/,10H J =,I2,/,10(3X,F8.1))
207 FORMAT(/,4H YL=,F10.4,4H YS=,F10.4)
212 FORMAT(/,4H VE=,E12.4,4H TB=,F12.2)
6 FORMAT(25HOVALUES FOR SN1(N) ARE/(1H ,5E14.4))
7 FORMAT(25HOVALUES FOR SN2(N) ARE/(1H ,5E14.4))
8 FORMAT(25HOVALUES FOR SN3(N) ARE/(1H ,5E14.4))

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PRI00460
PRI00470
PRI00480
PRI00490
PRI00500
PRI00510
PRI00520
PRI00530
PRI00540
PRI00550
PRI00560
PRI00570
PRI00580
PRI00590
PRI00600
PRI00610
PRI00620
PRI00630
PRI00640
PRI00650
PRI00660
PRI00670
PRI00680
PRI00690
PRI00700
PRI00710
PRI00720
PRI00730
PRI00740
PRI00750
PRI00760
PRI00770
PRI00780
PRI00790
PRI00800
PRI00810
PRI00820
PRI00830
PRI00840
PRI00850
PRI00860
PRI00870
PRI00880
PRI00890
PRI00900

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272

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15 FORMAT(25HOVALUES FOR YP1(N) ARE/(1H ,5E10.3)) PRI00910
16 FORMAT(25HOVALUES FOR YP2(J) ARE/(1H ,5E10.3)) PRI00920
17 FORMAT(25HOVALUES FOR YP3(N) ARE/(1H ,5E10.3)) PRI00930
19 FORMAT(25HOVALUES FOR YP5(J) ARE/(1H ,5E10.3)) PRI00940
END PRI00950
```





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DO 2 I=1,IN                                PR000460
DO 2 J=1,JN                                PR000470
IF(I.LE.I1.AND.J.LE.J1) GO TO 3            PR000480
IF(I.LT.IW(J).AND.J.LT.J1) GO TO 3        PR000490
IF(I.GT.I3) GO TO 3                        PR000500
A(I,J,NRO)=ROREF(2)                        PR000510
A(I,J,NMU)=ZMUREF(1)                       PR000520
A(I,J,NTC)=TCREF(2)                       PR000530
A(I,J,NSP)=SPREF(2)                       PR000540
GO TO 2                                    PR000550
3 A(I,J,NRO)=ROREF(1)                      PR000560
A(I,J,NTC)=TCREF(1)                       PR000570
A(I,J,NSP)=SPREF(1)                       PR000580
A(I,J,NS)=S(1)                             PR000590
A(I,J,NMU)=ZMUREF(1)                      PR000600
2 CONTINUE                                 PR000610
1 CONTINUE                                 PR000620
DO 10 I=I1,I3                              PR000630
DO 10 J=1,JN                              PR000640
IF(I.LT.IW(J).AND.J.LT.J1) GO TO 10        PR000650
A(I,J,NS)=S(2)                             PR000660
10 CONTINUE                                PR000670
IF(L.EQ.1.OR.L.EQ.2) GO TO 5               PR000680
C *** RECALCULATE VE,VC                    PR000690
HRM=HLEH                                    PR000700
HCON=SPREF(1)*(TLM-TA)+HL                  PR000710
AMELT=HRM/HCON                             PR000720
Z=VE                                        PR000730
VE=AMELT/(R(J1)*R(J1)*ROREF(1)*3.142)     PR000740
VC=VE*R(J1)*R(J1)/(R(JN)*R(JN))          PR000750
VC=VC*ROREF(1)/ROREF(5)                   PR000760
HCD=HCD*VE/Z                              PR000770
5 CONTINUE                                 PR000780
RETURN                                     PR000790
END                                         PR000800
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SUBROUTINE SORCE(SOURCE, I, J, K)	SOR00010
C\$	SOR00020
COMMON/CNUMB/NW, NF, NK, NEP, NV1, NV2, NT, NMU, NRO, NTC, NSP, NMT	SOR00030
COMMON/CCORD/IMIN(21), IMAX(21), X1(31), X2(21), R(21)	SOR00040
COMMON/CGRID/IN, INM, JN, JNM, I1, I2, I3, IA, IW(21), J1	SOR00050
COMMON/CDIM/X(5), RE, RM	SOR00060
COMMON/CITER/NMAX, NPRINT, NITER, IP, IE, IV, CC	SOR00070
COMMON/CCONT/C1, C2, C3, CD	SOR00080
COMMON/CRLAX/RP(10), RSDU(10), SAN(10)	SOR00090
COMMON/CEMP/WF, P, S(4)	SOR00100
COMMON/CHJ/NHR, NHI, NJRR, NJRI, NJZR, NJZI, NUJ, NS	SOR00110
COMMON/CTHERM/TCREF(5), SPREF(5), BETA(2), ES, EE, EM, SB, HL	SOR00120
COMMON/CTEMP/TLM, TSM, TLS, TA, TM, TW	SOR00130
COMMON/CTRVL/VE, VC	SOR00140
COMMON/CHEAT/HC, HW(4)	SOR00150
COMMON/CPROP/ROREF(5), ZMUREF(2), PR(10), GAMA	SOR00160
COMMON/CVF/FES(21), FEM(21), FSM, FSE	SOR00170
COMMON/CDROP/TAU, HCD, D, QS, QM	SOR00180
COMMON/CTAUW/TAUW1(20), TAUW2(15), TAUW3(21), TAUW4(10), TAUW5(21)	SOR00190
1, CAPPA, E	SOR00200
COMMON/CFLD/CU	SOR00210
COMMON/CPI/PI	SOR00220
COMMON/CTYP/LTYP	SOR00230
COMMON/CANG/THETA	SOR00240
COMMON/FITER/MIT, MPRINT, CP	SOR00250
COMMON/CDVAR/A(31, 21, 20)	SOR00260
COMMON/CNAME/ANAME, ASYMBL	SOR00270
COMMON/CCRIT/NWI, NWJ, NFI, NFJ, NTI, NTJ, NKI, NKJ, NEI, NEJ, GOSA, TOTA	SOR00280
COMMON/CLF/LF, LI	SOR00290
COMMON/CSVOR/SVOR(31, 21)	SOR00300
COMMON/CTURB/NTURB, NSCHEM, NVISC, RPXI, XI	SOR00310
COMMON/CSNW/SN1(20), SN2(15), SN3(21), SN4(20), SN5(21)	SOR00320
COMMON/CB/BE(31), BW(31), BN(21), BS(21)	SOR00330
COMMON/CAVM/AVM(60)	SOR00340
COMMON/CHBL/HLW, HLEH, HLEV, HLS, HLD, HLM, POWR	SOR00350
COMMON/CSUR/HS(2), TS(2)	SOR00360
COMMON/CBB/BBE, BBW, BBN, BBS, BPP	SOR00370
COMMON/CGEOM/FR1, FR2(15), FR3, FR4, FR5	SOR00380
COMMON/CAF/FUN(20)	SOR00390
COMMON/CVOLT/VOLT	SOR00400
COMMON/CAVT/TB	SOR00410
COMMON/CYP/YP1(20), YP2(15), YP3(21), YP4(10), YP5(21)	SOR00420
C\$	SOR00430
COMMON/CPROD/GK(31, 21)	SOR00440
COMMON/CSHEAR/GKK(31, 31)	SOR00450

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COMMON/CLREYN/LREYN,CDMOD(31,31)          SOR00460
C      BRANCH TO DIFFERENT SOURCE POINTS   SOR00470
C      -----                             SOR00480
C              GO TO (1,2,3,4,5,6,7,8,8),K SOR00490
C-----
C      DUMMY SOURCES                       SOR00500
C      -----                             SOR00510
C      8 SOURCE=0.0                         SOR00520
C      RETURN                               SOR00530
C-----
C      FOR RE(H)/R                         SOR00540
C      -----                             SOR00550
C      1 SOURCE=WF*P*A(I,J,NHI)           SOR00560
C      RETURN                               SOR00570
C      ..... FOR IM(H)/R                 SOR00580
C      -----                             SOR00590
C      2 SOURCE=-WF*P*A(I,J,NHR)         SOR00600
C      RETURN                               SOR00610
C-----
C      ..... FOR VORTICITY/R             SOR00620
C      -----                             SOR00630
C      3 SOURCE=A(I,J,NHR)*A(I,J,NJRR)+A(I,J,NHI)*A(I,J,NJRI) SOR00640
C      SOURCE=(1.0E+06)*P*R(J)*SOURCE    SOR00650
C      SVOR(I,J)=SOURCE                   SOR00660
C      RETURN                              SOR00670
C-----
C      ..... STREAM FUNCTION             SOR00680
C      -----                             SOR00690
C      4 SOURCE=A(I,J,NW)                 SOR00700
C      RETURN                              SOR00710
C-----
C      ..... TURBULENCE ENERGY         SOR00720
C      -----                             SOR00730
C      5 CONTINUE                         SOR00740
C      B1=ADF(I,J,1,NV1)                  SOR00750
C      B2=ADF(I,J,2,NV2)                  SOR00760
C      B3=ADF(I,J,2,NV1)                  SOR00770
C      B4=ADF(I,J,1,NV2)                  SOR00780
C      B5=A(I,J,NV2)/R(J)                 SOR00790
C      SHEAR2=(2.0*(B1**2+B2**2+B5**2)+(B3+B4)**2) SOR00800
C      GKK(I,J)=SHEAR2                    SOR00810
C      GK(I,J)=A(I,J,NMT) * SHEAR2       SOR00820
C      DK=A(I,J,NRO)*A(I,J,NEP)          SOR00830
C      SOURCE=GK(I,J)-DK                  SOR00840
C      RETURN                              SOR00850
C-----
C      ..... TURBULENCE ENERGY         SOR00860
C      -----                             SOR00870
C      5 CONTINUE                         SOR00880
C      B1=ADF(I,J,1,NV1)                  SOR00890
C      B2=ADF(I,J,2,NV2)                  SOR00900
C      B3=ADF(I,J,2,NV1)                  SOR00910
C      B4=ADF(I,J,1,NV2)                  SOR00920
C      B5=A(I,J,NV2)/R(J)                 SOR00930
C      SHEAR2=(2.0*(B1**2+B2**2+B5**2)+(B3+B4)**2) SOR00940
C      GKK(I,J)=SHEAR2                    SOR00950
C      GK(I,J)=A(I,J,NMT) * SHEAR2       SOR00960
C      DK=A(I,J,NRO)*A(I,J,NEP)          SOR00970
C      SOURCE=GK(I,J)-DK                  SOR00980
C      RETURN                              SOR00990

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C-----	SOR00910
C	SOR00920
C	SOR00930
6	SOR00940
CONTINUE	SOR00950
IF(A(I,J,NK).EQ.O.O) GO TO 100	SOR00960
TERM1=O.O	SOR00970
TERM2=C1*GK(I,J)*A(I,J,NEP)/A(I,J,NK)	SOR00980
F2=1.O	SOR00990
IF(LREYN.EQ.1) REYLOC=A(I,J,NRO)*A(I,J,NK)**2/(ZMUREF(2)	SOR01000
1*A(I,J,NEP))	SOR01010
IF(LREYN.EQ.1.AND.REYLOC.LE.10.O) F2=1.O -.3*EXP(-REYLOC**2)	SOR01020
C2MOD=C2 *F2	SOR01030
TERM3=C2MOD*A(I,J,NRO)*(A(I,J,NEP)**2)/A(I,J,NK)	SOR01040
SOURCE=TERM1+TERM2-TERM3	SOR01050
RETURN	SOR01060
100 SOURCE=O.O	SOR01070
RETURN	SOR01080
C-----	SOR01090
C	SOR01100
C	SOR01100
7	SOR01110
CONTINUE	SOR01120
IF(J.GT.J1) GO TO 9	SOR01130
Z=X1(I3)-X1(I2)	SOR01140
SOURCE=A(I,J,NJU)-QS/(3.14*R(J1)*R(J1)*Z)	SOR01150
GO TO 10	SOR01160
9 SOURCE=A(I,J,NJU)	SOR01170
10 CONTINUE	SOR01180
C-----	SOR01190
RETURN	SOR01200
END	



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	DO 22 I=1,I3	STRO0460
	DO 22 J=2,JNM	STRO0470
	IF(ABS(Z2).LT.ABS(A(I,J,NF))) Z2=A(I,J,NF)	STRO0480
22	CONTINUE	STRO0490
	DO 21 J=2,JNM	STRO0500
	IL=I1+1	STRO0510
	IF(J.LE.J1) IL=IW(J)+1	STRO0520
	IH=I3-1	STRO0530
	DO 21 I=IL,IH	STRO0540
	RISQ=1./R(J)/R(J)	STRO0550
	ROP=A(I,J,NRO)	STRO0560
	BBE=4./(A(I+1,J,NRO)+ROP)*RISQ*BE(I)	STRO0570
	BBW=4./(A(I-1,J,NRO)+ROP)*RISQ*BW(I)	STRO0580
	BBN=16./(A(I,J+1,NRO)+ROP)/((R(J+1)+R(J))**2)*BN(J)	STRO0590
	BBS=16./(A(I,J-1,NRO)+ROP)/((R(J-1)+R(J))**2)*BS(J)	STRO0600
	CALL SORCE(SOURCE,I,J,NF)	STRO0610
C	CALL CONVEC(AE,AW,AN,AS,I,J,NF)	STRO0620
	ANUM=BBE*A(I+1,J,NF)+BBW*A(I-1,J,NF)+BBN*A(I,J+1,NF)	STRO0630
	1+BBS*A(I,J-1,NF)+SOURCE	STRO0640
	ADNM=BBE+BBW+BBN+BBS	STRO0650
	IF(ADNM.EQ.O.) GO TO 5	STRO0660
	Z=A(I,J,NF)	STRO0670
	A(I,J,NF)=ANUM/ADNM	STRO0680
	IF(Z.EQ.O.O.AND.A(I,J,NF).EQ.O.O) GO TO 8	STRO0690
	RS=1.-Z/A(I,J,NF)	STRO0700
	GO TO 6	STRO0710
8	RS=O.O	STRO0720
6	CONTINUE	STRO0730
	GOSA=GOSA+ABS(Z-A(I,J,NF))	STRO0740
	TOTA=TOTA+ABS(A(I,J,NF))	STRO0750
	Z1=RP(NF)	STRO0760
	IF(J.GE.J1) Z1=O.2	STRO0770
	A(I,J,NF)=Z+Z1*(A(I,J,NF)-Z)	STRO0780
	IF(ABS(RS).LE.ABS(RSDU(NF))) GO TO 5	STRO0790
	RSDU(NF)=RS	STRO0800
	NFJ=J	STRO0810
	NFI=I	STRO0820
5	CONTINUE	STRO0830
21	CONTINUE	STRO0840
	RETURN	STRO0850
	END	STRO0860

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SUBROUTINE TURVAR(K)	TURO0010
C\$	TURO0020
COMMON/CNUMB/NW,NF,NK,NEP,NV1,NV2,NT,NMU,NRO,NTC,NSP,NMT	TURO0030
COMMON/CCORD/IMIN(21),IMAX(21),X1(31),X2(21),R(21)	TURO0040
COMMON/CGRID/IN,INM,JN,JNM,I1,I2,I3,IA,IW(21),J1	TURO0050
COMMON/CDIM/X(5),RE,RM	TURO0060
COMMON/CITER/NMAX,NPRINT,NITER,IP,IE,IV,CC	TURO0070
COMMON/CCONT/C1,C2,C3,CD	TURO0080
COMMON/CRLAX/RP(10),RSDU(10),SAN(10)	TURO0090
COMMON/CEMP/WF,P,S(4)	TURO0100
COMMON/CHJ/NHR,NHI,NJRR,NJRI,NJZR,NJZI,NJJ,NS	TURO0110
COMMON/CTHERM/TCREF(5),SPREF(5),BETA(2),ES,EE,EM,SB,HL	TURO0120
COMMON/CTEMP/TLM,TSM,TLS,TA,TM,TW	TURO0130
COMMON/CTRVL/VE,VC	TURO0140
COMMON/CHEAT/HC,HW(4)	TURO0150
COMMON/CPROP/ROREF(5),ZMUREF(2),PR(10),GAMA	TURO0160
COMMON/CVF/FES(21),FEM(21),FSM,FSE	TURO0170
COMMON/CDROP/TAU,HCD,D,QS,QM	TURO0180
COMMON/CTAUW/TAUW1(20),TAUW2(15),TAUW3(21),TAUW4(10),TAUW5(21)	TURO0190
1,CAPPA,E	TURO0200
COMMON/CFLD/CU	TURO0210
COMMON/CPI/PI	TURO0220
COMMON/CANG/THETA	TURO0230
COMMON/FITER/MIT,MPRINT,CP	TURO0240
COMMON/CDVAR/A(31,21,20)	TURO0250
COMMON/CNAME/ANAME,ASYMBL	TURO0260
COMMON/CCRIT/NWI,NWJ,NFI,NFJ,NTI,NTJ,NKI,NKJ,NEI,NEJ,GOSA,TOTA	TURO0270
COMMON/CLF/LF,LI	TURO0280
COMMON/CTYP/LTYP	TURO0290
COMMON/CSVOR/SVOR(31,21)	TURO0300
COMMON/CTURB/NTURB,NSCHEM,NVISC,RPXI,XI	TURO0310
COMMON/CSNW/SN1(20),SN2(15)	TURO0320
COMMON/CB/BE(31),BW(31),BN(21),BS(21)	TURO0330
COMMON/CAVM/AVM(60)	TURO0340
COMMON/CHBL/HLW,HLEH,HLEV,HLS,HLD,HLM,POWR	TURO0350
COMMON/CSUR/HS(2),TS(2)	TURO0360
COMMON/CBB/BBE,BBW,BBN,BBS,BPP	TURO0370
COMMON/CSORSE/SOURCE,SPRIME	TURO0380
COMMON/CGEOM/FR1,FR2(15),FR3,FR4,FR5	TURO0390
COMMON/CAF/FUN(20)	TURO0400
COMMON/CVOLT/VOLT	TURO0410
COMMON/CAVT/TB	TURO0420
COMMON/CYP/YP1(20),YP2(15),YP3(21),YP4(10),YP5(21)	TURO0430
C\$	TURO0440
COMMON/CPROD/GK(31,21)	TURO0450

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COMMON/CSHEAR/GKK(31,31)
DATA NBAL/O/
C
C.....FIRST GET THE CURRENT VALUE OF XI BASED ON OLD K.ENERGY FIELD.
C-----
                IF(NSCHEM.EQ.O) GO TO 800
                XIOLD=XI
                CALL CONLOT(XI,PERFIN,PCASC)
                XINEW=XI
                XI=XIOLD+RPXI*(XINEW-XIOLD)
                IF(XIOLD.EQ.O.O) XI=XINEW
C-----
                XI=050.00
C-----
C.....START GRID POINT CALCULATIONS
C-----
800                CONTINUE
                Z2=0.
                DO 23 I=1,I3
                DO 23 J=2,JNM
                IF(ABS(Z2).LT.ABS(A(I,J,K))) Z2=A(I,J,K)
23                CONTINUE
                                ENSUM=0.0
                                GENSUM=0.0
                                DISSUM=0.0
                DO 31 J=2,JNM
                IL=I+1
                IF(J.LE.J1) IL=IW(J)+1
                IH=I3-1
                DO 31 I=IL,IH
                BPP=A(I,J,NMU)
                BBE=(A(I+1,J,NMU)+BPP)/PR(K)*BE(I)
                BBW=(A(I-1,J,NMU)+BPP)/PR(K)*BW(I)
                BBN=(A(I,J+1,NMU)+BPP)/PR(K)*BN(J)
                BBS=(A(I,J-1,NMU)+BPP)/PR(K)*BS(J)
36                SPRIME=0.
                CALL SORCE(SOURCE,I,J,K)
                CALL CONVEC(AE,AW,AN,AS,I,J,K)
                ANUM=(AE+BBE)*A(I+1,J,K)+(AW+BBW)*A(I-1,J,K)+(AN+BBN)*A(I,J+1,K)
                +((AS+BBS)*A(I,J-1,K)+SOURCE
                ADN=AE+AW+AN+AS+BBE+BBW+BBN+BBS+SPRIME
                IF(ADNM.EQ.O.) GO TO 6
                Z=A(I,J,K)
                A(I,J,K)=ANUM/ADNM
                IF(Z.EQ.O.O.AND.A(I,J,K).EQ.O.O) GO TO 3

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TURO0460
TURO0470
TURO0480
TURO0490
TURO0500
TURO0510
TURO0520
TURO0530
TURO0540
TURO0550
TURO0560
TURO0570
TURO0580
TURO0590
TURO0600
TURO0610
TURO0620
TURO0630
TURO0640
TURO0650
TURO0660
TURO0670
TURO0680
TURO0690
TURO0700
TURO0710
TURO0720
TURO0730
TURO0740
TURO0750
TURO0760
TURO0770
TURO0780
TURO0790
TURO0800
TURO0810
TURO0820
TURO0830
TURO0840
TURO0850
TURO0860
TURO0870
TURO0880
TURO0890
TURO0900

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      RS=1.-Z/A(I,J,K)
      GO TO 5
3     RS=0.0
5     CONTINUE
      GOSA=GOSA+ABS(Z-A(I,J,K))
      TOTA=TOTA+ABS(A(I,J,K))
      Z1=RP(K)
C     IF(J.GE.10) Z1=0.2
      A(I,J,K)=Z+RP(K)*(A(I,J,K)-Z)
      IF(A(I,J,K).LE.0.0) A(I,J,K)=0.0
      IF(NBAL.EQ.0) GO TO 100
C..... CALCULATE ENERGY BUDGET
C-----
C     SPOINT=SOURCE-SPRIME*A(I,J,NK)
C     DR=(R(J+1)-R(J-1))/2.0
C     DX=(X1(I+1)-X1(I-1))/2.0
C     ENSUM=ENSUM+SPOINT*R(J)*DR*DX
C     GENSUM=GENSUM+GK(I,J)*R(J)*DR*DX
C     DISSUM=DISSUM+A(I,J,NEP)*A(I,J,NRO)*R(J)*DR*DX
C-----
100    CONTINUE
      IF(ABS(RS).LT.ABS(RSDU(K))) GO TO 6
      RSDU(K)=RS
      IF(K.EQ.NEP) GO TO 7
      NKI=I
      NKJ=J
      GO TO 6
7     NEI=I
      NEJ=J
6     CONTINUE
31    CONTINUE
      IF(NBAL.EQ.0) RETURN
C     CALL VOL(GK,GENVOL)
C... PRINT OUT THE ENERGY BUDGET
C-----
C     IF(K.EQ.NK)WRITE(6,200) ENSUM,GENSUM,DISSUM,GENVOL
C200  FORMAT('ENERGY BALANCE= ',1PE10.2)
C     1/'GENERATION = ',1PE10.2
C     2/'DISSIPATION= ',1PE10.2
C     3/'GENERATION #2 ',1PE10.2)
C-----
      RETURN
      END

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TURO0910
TURO0920
TURO0930
TURO0940
TURO0950
TURO0960
TURO0970
TURO0980
TURO0990
TURO1000
TURO1010
TURO1020
TURO1030
TURO1040
TURO1050
TURO1060
TURO1070
TURO1080
TURO1090
TURO1100
TURO1110
TURO1120
TURO1130
TURO1140
TURO1150
TURO1160
TURO1170
TURO1180
TURO1190
TURO1200
TURO1210
TURO1220
TURO1230
TURO1240
TURO1250
TURO1260
TURO1270
TURO1280
TURO1290
TURO1300
TURO1310
TURO1320
TURO1330

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      IL=I1+1
      IF(J.LE.J1) IL=IW(J)+1
      IH=I3-1
      DO 51 I=IL,IH
      A(I,J,NV1)=ADF(I,J,2,NF)/R(J)/A(I,J,NRO)
      A(I,J,NV2)=-ADF(I,J,1,NF)/R(J)/A(I,J,NRO)
51  CONTINUE
50  CONTINUE
      IL=IW(1)+1
      IH=I3-1
      RR=R(2)*R(2)/(R(3)*R(3))
      DO 54 I=IL,IH
      A(I,1,NV1)=2.0*(A(I,2,NF)-RR*RR*A(I,3,NF))
      1/(R(2)*R(2)*(1.-RR)*A(I,1,NRO))
54  CONTINUE
      J12=J1+1
      DO 40 J=J12,JNM
      A(I1,J,NV2)=-ADF(I1,J,1,NF)/R(J)/ROREF(2)
40  CONTINUE
      RETURN
      END
```

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VEL00460
VEL00470
VEL00480
VEL00490
VEL00500
VEL00510
VEL00520
VEL00530
VEL00540
VEL00550
VEL00560
VEL00570
VEL00580
VEL00590
VEL00600
VEL00610
VEL00620
VEL00630
VEL00640
VEL00650
VEL00660
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SUBROUTINE VISCOS	VIS00010
C\$	VIS00020
COMMON/CNUMB/NW,NF,NK,NEP,NV1,NV2,NT,NMU,NRO,NTC,NSP,NMT	VIS00030
COMMON/CCORD/IMIN(21),IMAX(21),X1(31),X2(21),R(21)	VIS00040
COMMON/CGRID/IN,INM,JN,JNM,I1,I2,I3,IA,IW(21),J1	VIS00050
COMMON/CDIM/X(5),RE,RM	VIS00060
COMMON/CITER/NMAX,NPRINT,NITER,IP,IE,IV,CC	VIS00070
COMMON/CCONT/C1,C2,C3,CD	VIS00080
COMMON/CRLAX/RP(10),RSDU(10),SAN(10)	VIS00090
COMMON/CEMP/WF,P,S(4)	VIS00100
COMMON/CHJ/NHR,NHI,NJRR,NJRI,NJZR,NJZI,NJJ,NS	VIS00110
COMMON/CTHERM/TCREF(5),SPREF(5),BETA(2),ES,EE,EM,SB,HL	VIS00120
COMMON/CTEMP/TLM,TSM,TLS,TA,TM,TW	VIS00130
COMMON/CTRVL/VE,VC	VIS00140
COMMON/CHEAT/HC,HW(4)	VIS00150
COMMON/CPROR/ROREF(5),ZMUREF(2),PR(10),GAMA	VIS00160
COMMON/CVF/FES(21),FEM(21),FSM,FSE	VIS00170
COMMON/CDROP/TAU,HCD,D,QS,QM	VIS00180
COMMON/CTAUW/TAUW1(20),TAUW2(15),TAUW3(21),TAUW4(10),TAUW5(21)	VIS00190
1,CAPPA,E	VIS00200
COMMON/CFLD/CU	VIS00210
COMMON/CPI/PI	VIS00220
COMMON/CANG/THETA	VIS00230
COMMON/FITER/MIT,MPRINT,CP	VIS00240
COMMON/CDVAR/A(31,21,20)	VIS00250
COMMON/CNAME/ANAME,ASYMBL	VIS00260
COMMON/CCRIT/NWI,NWJ,NFI,NFJ,NTI,NTJ,NKI,NKJ,NEI,NEJ,GOSA,TOTA	VIS00270
COMMON/CLF/LF,LI	VIS00280
COMMON/CTYP/LTYP	VIS00290
COMMON/CSVOR/SVOR(31,21)	VIS00300
COMMON/CTURB/NTURB,NSCHEM,NVISC,RPXI,XI	VIS00310
COMMON/CSNW/SN1(20),SN2(15)	VIS00320
COMMON/CB/BE(31),BW(31),BN(21),BS(21)	VIS00330
COMMON/CAVM/AVM(60)	VIS00340
COMMON/CHBL/HLW,HLEH,HLEV,HLS,HLD,HLM,POWR	VIS00350
COMMON/CSUR/HS(2),TS(2)	VIS00360
COMMON/CBB/BBE,BBW,BBN,BBS,BPP	VIS00370
COMMON/CSORSE/SOURCE,SPRIME	VIS00380
COMMON/CGEOM/FR1,FR2(15),FR3,FR4,FR5	VIS00390
COMMON/CAF/FUN(20)	VIS00400
COMMON/CVOLT/VOLT	VIS00410
COMMON/CAVT/TB	VIS00420
COMMON/CYP/YP1(20),YP2(15),YP3(21),YP4(10),YP5(21)	VIS00430
C\$	VIS00440
COMMON/CLREYN/LREYN,CDMOD(31,31)	VIS00450

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C*** CALCULATE TURBULENT VISCOSITY          VIS00460
  I31=I3-1                                   VIS00470
  DO 10 J=1,JNM                               VIS00480
  DO 11 I=I1,I31                             VIS00490
  IF(J.LE.J1.AND.I.LE.IW(J)) GO TO 12       VIS00500
  IF(A(I,J,NEP).LE.O.O.OR.A(I,J,NK).LE.O.O) VIS00510
  A(I,J,NMT) =A(I,J,NRO)*CDMOD(I,J)*(A(I,J,NK)**2)/A(I,J,NEP) VIS00520
  GO TO 21                                    VIS00530
20 CONTINUE                                  VIS00540
  A(I,J,NMT)=O.                               VIS00550
21 CONTINUE                                  VIS00560
  Z=A(I,J,NMU)                                VIS00570
  EMU=ZMUREF(1)                              VIS00580
  A(I,J,NMU)=EMU +A(I,J,NMT)                 VIS00590
  A(I,J,NMU)=Z+RP(NMU)*(A(I,J,NMU)-Z)       VIS00600
12 CONTINUE                                  VIS00610
11 CONTINUE                                  VIS00620
10 CONTINUE                                  VIS00630
  RETURN                                       VIS00640
  END                                          VIS00650
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C      TO CALCULATE THE LOTSIANSKII CONSTANT
      VAL=0.0
      DO 100 J=2,JNM
      IL=IW(J)+1
      DO 100 I=IL,INM
      DR=(R(J+1)-R(J-1))/2.0
      DX=(X1(I+1)-X1(I-1))/2.0
      DV=R(J)*DR*DX
      IF(I.EQ.IL.OR.I.EQ.IH)DV=1.5*DV
      IF(J.EQ.JNM) DV=1.5*DV
      VAL=VAL+P(I,J)*DV
100    CONTINUE
      DO 150 I=IL,IH
      VAL=VAL + P(I,1)*R(1)**2*(X1(I+1)-X1(I-1))/4.0
150    CONTINUE
      RETURN
      END
```

VOL00460  
VOL00470  
VOL00480  
VOL00490  
VOL00500  
VOL00510  
VOL00520  
VOL00530  
VOL00540  
VOL00550  
VOL00560  
VOL00570  
VOL00580  
VOL00590  
VOL00600  
VOL00610  
VOL00620



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SUBROUTINE VORITY
COMMON/CB/BE(31),BW(31),BN(21),BS(21)
COMMON/CDVAR/A(31,21,20)
COMMON/CNUMB/NW,NF,NK,NEP,NV1,NV2,NT,NMU,NRO,NTC,NSP,NMT
COMMON/CCORD/IMIN(21),IMAX(21),X1(31),X2(21),R(21)
COMMON/CCRIT/NWI,NWJ,NFI,NFJ,NTI,NTJ,NKI,NKJ,NEI,NEJ,GOSA,TOTA
COMMON/CRLAX/RP(10),RSDU(10),SAN(10)
COMMON/CGRID/IN,INM,JN,JNM,I1,I2,I3,IA,IW(21),J1
COMMON/CTHERM/TCREF(5),SPREF(5),BETA(2),ES,EE,EM,SB,HL
COMMON/CBB/BBE,BBW,BBN,BBS,BPP
COMMON/CSORSE/SOURCE,SPRIME
C-----
C          SOLUTION OF VORTICITY TRANSPORT EQUATION
C-----
      IL=I1+1
C      IF(J.LE.J1) IL=IW(J)+1
      IH=I3-1
      DO 11 I=IL,IH
      DO 11 J=2,JNM
      RSQ=R(J)*R(J)
      BBE=2.*RSQ*BE(I)
      BBW=2.*RSQ*BW(I)
      BBN=(R(J+1)*R(J+1)+RSQ)*BN(J)
      BBS=(R(J-1)*R(J-1)+RSQ)*BS(J)
      SOURCE=0.0
      CALL CONVEC(AE,AW,AN,AS,I,J,NW)
C.....          IMPLICIT VORTICITY CALCULATIONS
      TERM1=A(I+1,J,NW)
      TERM2=A(I-1,J,NW)
      TERM3=A(I,J-1,NW)
      TERM4=A(I,J+1,NW)
      TERM5=0.0
      IF(I.NE.(I1+1)) GO TO 14
      DX12=(X1(I1+1)-X1(I1))**2
C.....          ALLOW FOR COMPLETE INFLOW WALL CONDITIONS
      TERM2=TERM2+
13.*(A(I1,J,NF)-A(I1+1,J,NF))/(DX12*RSQ*A(I1,J,NRO))
      TERM5=0.5*(AW+A(I-1,J,NMU)*BBW)
C.....          CALCULATE NEW VORTICITY.....
14      CONTINUE
      ANUM=(AE+A(I+1,J,NMU)*BBE)*TERM1
      1+(AW+A(I-1,J,NMU)*BBW)*TERM2
      2+(AS+A(I,J-1,NMU)*BBS)*TERM3
      3+(AN+A(I,J+1,NMU)*BBN)*TERM4          +SOURCE
      ADN=AE+AW+AN+AS+A(I,J,NMU)*(BBE+BBW+BBN+BBS)+TERM5

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VGRO0010
VORO0020
VORO0030
VORO0040
VORO0050
VORO0060
VORO0070
VORO0080
VORO0090
VORO0100
VORO0110
VORO0120
VORO0130
VORO0140
VORO0150
VORO0160
VORO0170
VORO0180
VORO0190
VORO0200
VORO0210
VORO0220
VORO0230
VORO0240
VORO0250
VORO0260
VORO0270
VORO0280
VORO0290
VORO0300
VORO0310
VORO0320
VORO0330
VORO0340
VORO0350
VORO0360
VORO0370
VORO0380
VORO0390
VORO0400
VORO0410
VORO0420
VORO0430
VORO0440
VORO0450

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IF(ADNM.EQ.O.) GO TO 4	VOR00460
Z=A(I,J,NW)	VOR00470
A(I,J,NW)=ANUM/ADNM	VOR00480
C.....RELAXATION AND RESIDUE CALCULATIONS	VOR00490
IF(Z.EQ.O.O.AND.A(I,J,NW).EQ.O.O) GO TO 1	VOR00500
RS=1.-Z/A(I,J,NW)	VOR00510
GO TO 3	VOR00520
1 RS=O.	VOR00530
3 CONTINUE	VOR00540
GOSA=GOSA+ABS(Z-A(I,J,NW))	VOR00550
TOTA=TOTA+ABS(A(I,J,NW))	VOR00560
C.....	VOR00570
Z1=RP(NW)	VOR00580
C IF(J.GE.J1) Z1=O.1	VOR00590
A(I,J,NW)=Z+Z1*(A(I,J,NW)-Z)	VOR00600
C.....	VOR00610
IF(ABS(RS).LE.ABS(RSDU(NW))) GO TO 4	VOR00620
RSDU(NW)=RS	VOR00630
NWJ=J	VOR00640
NWI=I	VOR00650
4 CONTINUE	VOR00660
C.....DIAGNOSTIC AIDS.....	VOR00670
WRITE(15,997) AE,AW,AN,AS,TERM1,TERM2,TERM3,TERM4	VOR00680
997 FORMAT(8(1PE10.2))	VOR00690
WRITE(15,998)I,J,A(I,J,NW)	VOR00700
998 FORMAT(2I5,1PE10.2)	VOR00710
WRITE(15,998) I1,J,A(I1,J,NF)	VOR00720
C.....	VOR00730
11 CONTINUE	VOR00740
RETURN	VOR00750
END	VOR00760

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SUBROUTINE WALL	WAL00010
\$	WAL00020
COMMON/CNUMB/NW,NF,NK,NEP,NV1,NV2,NT,NMU,NRO,NTC,NSP,NMT	WAL00030
COMMON/CCORD/IMIN(21),IMAX(21),X1(31),X2(21),R(21)	WAL00040
COMMON/CGRID/IN,INM,JN,UNM,I1,I2,I3,IA,IW(21),J1	WAL00050
COMMON/CDIM/X(5),RE,RM	WAL00060
COMMON/CITER/NMAX,NPRINT,NITER,IP,IE,IV,CC	WAL00070
COMMON/CCONT/C1,C2,C3,CD	WAL00080
COMMON/CRLAX/RP(10),RSDU(10),SAN(10)	WAL00090
COMMON/CEMP/WF,P,S(4)	WAL00100
COMMON/CHJ/NHR,NHI,NJRR,NJRI,NJZR,NJZI,NJJ,NS	WAL00110
COMMON/CTHERM/TCREF(5),SPREF(5),BETA(2),ES,EE,EM,SB,HL	WAL00120
COMMON/CTEMP/TLM,TSM,TLS,TA,TM,TW	WAL00130
COMMON/CTRVL/VE,VC	WAL00140
COMMON/CHEAT/HC,HW(4)	WAL00150
COMMON/CPROP/ROREF(5),ZMUREF(2),PR(10),GAMA	WAL00160
COMMON/CVF/FES(21),FEM(21),FSM,FSE	WAL00170
COMMON/CDROP/TAU,HCD,D,QS,QM	WAL00180
COMMON/CTAUW/TAUW1(20),TAUW2(15),TAUW3(21),TAUW4(10),TAUW5(21)	WAL00190
1,CAPPA,E	WAL00200
COMMON/CFLD/CU	WAL00210
COMMON/CPI/PI	WAL00220
COMMON/CTYP/LTYP	WAL00230
COMMON/CANG/THETA	WAL00240
COMMON/FITER/MIT,MPRINT,CP	WAL00250
COMMON/CDVAR/A(31,21,20)	WAL00260
COMMON/CNAME/ANAME,ASYMBL	WAL00270
COMMON/CCRIT/NWI,NWJ,NFI,NFJ,NTI,NTJ,NKI,NKJ,NEI,NEJ,GOSA,TOTA	WAL00280
COMMON/CLF/LF,LI	WAL00290
COMMON/CSVOR/SVOR(31,21)	WAL00300
COMMON/CTURB/NTURB,NSCHEM,NVISC,RPXI,XI	WAL00310
COMMON/CSNW/SN1(20),SN2(15),SN3(21),SN4(20),SN5(21)	WAL00320
COMMON/CB/BE(31),BW(31),BN(21),BS(21)	WAL00330
COMMON/CAVM/AVM(60)	WAL00340
COMMON/CHBL/HLW,HLEH,HLEV,HLS,HLD,HLM,POWR	WAL00350
COMMON/CSUR/HS(2),TS(2)	WAL00360
COMMON/CBB/BBE,BBW,BBN,BBS,BPP	WAL00370
COMMON/CSORSE/SOURCE,SPRIME	WAL00380
COMMON/CGEOM/FR1,FR2(15),FR3,FR4,FR5	WAL00390
COMMON/CAF/FUN(20)	WAL00400
COMMON/CVOLT/VOLT	WAL00410
COMMON/CAVT/TB	WAL00420
COMMON/CYP/YP1(20),YP2(15),YP3(21),YP4(10),YP5(21)	WAL00430
\$	WAL00440
COMMON/CLREYN/LREYN,CDMOD(31,31)	WAL00450

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PRL=SPREF(2)*ZMUREF(1)/TCREF(2)
RECPRT=1./PR(NT)
PRRAT=PRL*RECPRT
PJAY=9.*(PRRAT-1.)/(PRRAT**0.25)
ANU=ZMUREF(1)/ROREF(2)
C ***** AT J=JN,TAUW1 , SN1
YP=X2(JN)-X2(JNM)
ZL=X1(I3)-X1(I1)
IL=I1+1
IH=I3-1
DO 1 I=IL,IH
N=I-IL+1
A1=ROREF(2)*CDMOD(I,J)**0.25
A2=A1*SQRT(A(I,JNM,NK))
YPP=YP*A2/ZMUREF(1)
UP=ABS(A(I,JNM,NV1))
IF(YPP.GT.11.5) GO TO 2
TAUW1(N)=ZMUREF(1)*UP/YP
GO TO 3
2 TAUW1(N)=CAPPA*UP*A2/ALOG(YPP*E)
3 CONTINUE
C TAUGR=ROREF(2)*BETA(1)*9.81*(A(I,JNM,NT)-A(I,JN,NT))*YP/2.
TAUGR=0.
TAUEM=A(I,JNM,NJRR)*A(I,JNM,NHR)+A(I,JNM,NJRI)*A(I,JNM,NHI)
TAUEM=TAUEM*(1.OE+06)*P*R(J)*YP/4.
PN=0.
C IF(A(I,JNM,NV1).LT.0.) PN=-1.
TAUW1(N)=ABS(TAUW1(N))+ABS(TAUGR)+ABS(TAUEM)*PN
TAUW1(N)=ABS(TAUW1(N))
FRIC=TAUW1(N)/(UP*UP*ROREF(2))
16 CONTINUE
TCS=TCREF(2)+0.5*SPREF(2)*A(I,JNM,NMT)
DTG=A(I,JNM,NJJ)*YP*YP/(2.*TCS)
F1=1.
STAN=FRIC*RECPRT/(1.+PJAY*SQRT(FRIC))
SN1(N)=ABS(STAN)*ROREF(2)*SPREF(2)*UP*F1
YP1(N)=YPP
1 CONTINUE
C *** AT I=IW+1,1<J<J1
DO 4 J=2,J1
II=IW(J)
YP=(X1(II+1)-X1(II))*COS(THETA)
A1=ROREF(2)*CDMOD(I,J)**0.25
A2=A1*SQRT(A(II+1,J,NK))
YPP=YP*A2/ZMUREF(1)

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WAL00460
WAL00470
WAL00480
WAL00490
WAL00500
WAL00510
WAL00520
WAL00530
WAL00540
WAL00550
WAL00560
WAL00570
WAL00580
WAL00590
WAL00600
WAL00610
WAL00620
WAL00630
WAL00640
WAL00650
WAL00660
WAL00670
WAL00680
WAL00690
WAL00700
WAL00710
WAL00720
WAL00730
WAL00740
WAL00750
WAL00760
WAL00770
WAL00780
WAL00790
WAL00800
WAL00810
WAL00820
WAL00830
WAL00840
WAL00850
WAL00860
WAL00870
WAL00880
WAL00890
WAL00900

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UP1=A(II+1,J,NV2)*COS(THETA)-A(II+1,J,NV1)*SIN(THETA)
UP=ABS(UP1)
IF(YPP.GT.11.5) GO TO 5
TAUW2(J)=ZMUREF(1)*UP/YP
GO TO 6
5 TAUW2(J)=CAPPA*UP*A2/ALOG(YPP*E)
6 CONTINUE
TAUE1=A(II+1,J,NJZR)*A(II+1,J,NHR)+A(II+1,J,NJZI)*A(II+1,
1J,NHI)
TAUE2=A(II+1,J,NJRR)*A(II+1,J,NHR)+A(II+1,J,NJRI)*A(II+1,
1J,NHI)
TAUE=TAUE1*COS(THETA)+TAUE2*SIN(THETA)
TAUEM=ABS(TAUE)
TAUEM=TAUEM*(1.OE+06)*P*R(J)*YP/4.
PN=0.
C IF(UP1.GT.0.) PN=-1.
TAUW2(J)=ABS(TAUW2(J))+ABS(TAUEM)*PN
TAUW2(J)=ABS(TAUW2(J))
FRIC=TAUW2(J)/(UP*UP*RREF(2))
TCS=TCREF(2)+0.5*SPREF(2)*A(I2+1,J,NMT)
DTG=A(I2+1,J,NJJ)*YP*YP/(2.*TCS)
F2=1.
STAN=FRIC*RECPRT/(1.+PJAY*SQRT(FRIC))
SN2(J)=ABS(STAN)*RREF(2)*SPREF(2)*UP*F2
YP2(J)=YPP
40 CONTINUE
4 CONTINUE
C *** AT I=I3-1,1<J<JN , TAUW3 ,SN3
YP=X1(I3)-X1(I3-1)
DO 7 J=2,JNM
A2=A1*SQRT(A(I3-1,J,NK))
YPP=YP*A2/ZMUREF(1)
UP=ABS(A(I3-1,J,NV2))
IF(YPP.GT.11.5) GO TO 8
TAUW3(J)=ZMUREF(1)*UP/YP
GO TO 9
8 TAUW3(J)=CAPPA*UP*A2/ALOG(YPP*E)
9 CONTINUE
TAUEM=A(I3-1,J,NJZR)*A(I3-1,J,NHR)+A(I3-1,J,NJZI)*A(I3-1,
1J,NHI)
TAUEM=TAUEM*(1.OE+06)*P*R(J)*YP/4.
PN=0.
C IF(A(I3-1,J,NV2).GT.0.) PN=-1.
TAUW3(J)=ABS(TAUW3(J))+ABS(TAUEM)*PN
TAUW3(J)=ABS(TAUW3(J))

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WALO0910
WALO0920
WALO0930
WALO0940
WALO0950
WALO0960
WALO0970
WALO0980
WALO0990
WALO1000
WALO1010
WALO1020
WALO1030
WALO1040
WALO1050
WALO1060
WALO1070
WALO1080
WALO1090
WALO1100
WALO1110
WALO1120
WALO1130
WALO1140
WALO1150
WALO1160
WALO1170
WALO1180
WALO1190
WALO1200
WALO1210
WALO1220
WALO1230
WALO1240
WALO1250
WALO1260
WALO1270
WALO1280
WALO1290
WALO1300
WALO1310
WALO1320
WALO1330
WALO1340
WALO1350

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FRIC=TAUW3(J)/(UP*UP*RREF(2))
DT=A(I3-1,J,NT)-A(I3,J,NT)
IF(DT.LE.O.) DT=0.
18 CONTINUE
TCS=TCREF(2)+A(I3-1,J,NMT)*SPREF(2)*0.5
DTG=A(I3-1,J,NJJ)*YP*YP/(2.*TCS)
F3=1.
STAN=FRIC*RECPRT/(1.+PJAY*SQRT(FRIC))
SN3(J)=ABS(STAN)*RREF(2)*SPREF(2)*UP*F3
YP3(J)=YPP
7 CONTINUE
C *** AT J=J1+1,I1<I<I2
IF(I1.EQ.I2) GO TO 15
YP=X2(J1+1)-X2(J1)
IL=I1+1
I21=I2-1
DO 10 I=IL,I2
N=I-IL+1
A1=RREF(2)*CDMOD(I,J)**0.25
A2=A1*SQRT(A(I,J1+1,NK))
YPP=YP*A2/ZMUREF(1)
UP=ABS(A(I,J1+1,NV1))
IF(YPP.GT.11.5) GO TO 11
TAUW4(N)=ZMUREF(1)*UP/YP
GO TO 12
11 TAUW4(N)=CAPPA*UP*A2/ALOG(YPP*E)
12 CONTINUE
TAUEM=A(I,J1+1,NJRR)*A(I,J1+1,NHR)+A(I,J1+1,NJRI)*A(I,J1+
11,NHI)
TAUEM=TAUEM*(1.OE+06)*P*R(J)*YP/4.
PN=0.
TAUW4(N)=ABS(TAUW4(N))+ABS(TAUEM)*PN
FRIC=TAUW4(N)/(UP*UP*RREF(2))
STAN=FRIC*RECPRT/(1.+PJAY*SQRT(FRIC))
SN4(N)=ABS(STAN)*RREF(2)*SPREF(2)*UP
YP4(N)=YPP
10 CONTINUE
15 CONTINUE
20 CONTINUE
RETURN
END
WALO1360
WALO1370
WALO1380
WALO1390
WALO1400
WALO1410
WALO1420
WALO1430
WALO1440
WALO1450
WALO1460
WALO1470
WALO1480
WALO1490
WALO1500
WALO1510
WALO1520
WALO1530
WALO1540
WALO1550
WALO1560
WALO1570
WALO1580
WALO1590
WALO1600
WALO1610
WALO1620
WALO1630
WALO1640
WALO1650
WALO1660
WALO1670
WALO1680
WALO1690
WALO1700
WALO1710
WALO1720
WALO1730
WALO1740
WALO1750
WALO1760

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## NOMENCLATURE

a	Radius of heated cylinder (m)
B	Magnetic Flux Intensity (Tesla)
C <sub>1</sub> , C <sub>2</sub> , C <sub>d</sub>	Constants in the $\kappa - \epsilon$ Turbulence Model
D <sub>c</sub>	Diffusivity of mixing species in the melt (m <sup>2</sup> /sec)
D	Depth of melt (m)
E(k)	3-Dimensional Turbulence Energy Density Spectrum
E	Electric Field
f(r)	Spatial Autocorrelation Function for turbulence
F <sub>be</sub>	Electromagnetic Force field
g	Local Turbulence Energy generation term
G	Total Turbulence Energy generation over entire volume of melt
H	Magnetic Induction (Amps/m)
I	Total current (Amps)
I <sub>L</sub>	Lotsianskii Integral in XI Model of Turbulence
J	Current Density (Amps/m <sup>2</sup> )
k	Wavenumber (m <sup>-1</sup> )
k <sub>e</sub>	Wavenumber of Energy Containing Eddies
l	Energy Containing length scale
L <sub>v</sub>	Characteristic Vessel Dimension (m)
L <sub>p</sub>	Characteristic size of Large Scale Eddies (m)
M	Mass of melt

N	Magnetic Interaction Parameter
p	pressure in the fluid
P	Stirring Power input to the system (Watts)
$P_t$	Specific Power input (Watts/Ton of Melt)
PI	Stands for Performance Index used to evaluate degree of Turbulence
q	RMS Turbulent Fluctuating Velocity
$R_e$	Radius of electrode (m)
$R_m$	Radius of vessel (m)
r	Radial location in the melt (m)
t	time (secs)
$t_m$	Mixing time for 95% homogenisation (sec)
T	Temperature
U	Mean Velocity (m/sec)
$u'_{i,j}$	Fluctuating velocity components
V	Mean Nondimensional Velocity
x	Axial Location in melt (m)
$\lambda$	Taylor Microscale
$l$	Length scale for concentration fluctuations corresponding to $\lambda$
$\phi$	Turbulence Dissipation rate per unit mass (W/Kg-sec)
$\Phi$	Total Turbulence Dissipation Rate (W/sec)
$\sigma$	Electrical Conductivity (mho-m)
$\rho$	Density of the fluid (Kg/m <sup>3</sup> )



$\mu_o$	Magnetic Permeability of Free Space
$\eta$	Magnetic Diffusivity, Efficiency of Turbulence Generation
$\nu$	Molecular Kinematic Viscosity of Fluid ( $m^2 / sec$ )
$\mu$	Molecular Viscosity of Fluid (Kg./m-sec)
$\alpha$	Thermal Diffusivity of Fluid ( $m^2 / sec$ )
$\omega$	Vorticity
$\psi$	Stream Function
$\mu_{eff}$	Turbulence Enhanced Viscosity (Kg/m-sec)
$\mu_t$	Turbulent Contribution to Viscosity (Kg/m-sec)
$\kappa$	Kinetic Energy of Turbulent Fluctuations (W/Kg)
$\epsilon$	Local Rate of Turbulence Energy Dissipation (W/Kg-sec)
$\sigma_e$	'Prandtl Number' for $\epsilon$ Transport
$\sigma_k$	'Prandtl Number' for $\kappa$ Transport
$\xi$	Primary Variable in the XI Model of Turbulence
$\tau$	Time constant for Mixing (66% mixed)
$\tau$	Shear Stress Tensor

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