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Pressure Distributions From High Reynolds Number Tests of a Boeing BAC I Airfoil in the Langley 0.3-Meter Transonic Cryogenic Tunnel

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William G. Johnson, Jr., and Acquilla S. Hill

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William G. Johnson, Jr., and Acquilla S. Hill
Langley Research Center
Hampton, Virginia



National Aeronautics
and Space Administration

**Scientific and Technical
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Summary

In a cooperative effort with U.S. manufacturers of large transport aircraft, NASA has completed a systematic study of well-known conventional and advanced-technology airfoil concepts over a wide range of Reynolds number. This study, referred to as the Advanced Technology Airfoil Tests (ATAT) program, was conducted in the two-dimensional test section of the Langley 0.3-Meter Transonic Cryogenic Tunnel.

The pressure distributions presented in this report are from the first of several NASA/U.S. industry airfoil investigations conducted as part of the ATAT program. The industry participant for this investigation was the Boeing Commercial Airplane Company. The test was conducted on a Boeing 10-percent-thick airfoil which was designated as BAC I for the purposes of this test. Test temperature was varied from ambient to about 100 K at pressures ranging from about 122 to 608 kPa (1.2 to 6 atm). Mach number was varied from 0.40 to 0.80. These variables provided a Reynolds number range (based on airfoil model chord) from 4.4×10^6 to 50.0×10^6 .

The pressure data are presented without analysis in tabulated format and as plots of pressure coefficient as a function of position on the airfoil. This report was prepared for use in conjunction with the aerodynamic coefficient data published in NASA Technical Memorandum 81922.

Introduction

The National Aeronautics and Space Administration (NASA) has completed a systematic study of well-known conventional and advanced-technology airfoil concepts over a wide range of Reynolds number. This study, described in detail in reference 1, is referred to as the Advanced Technology Airfoil Tests (ATAT) program and was conducted in the Langley 0.3-Meter Transonic Cryogenic Tunnel (0.3-m TCT). References 2 through 26 report some of the data acquired during the ATAT program. As can be seen from these references, a significant portion of the advanced-technology airfoil phase of the ATAT program was carried out in cooperation with U.S. industry. Three U.S. manufacturers of large commercial transport aircraft (Boeing Commercial Airplane Company, Douglas Aircraft Company, and Lockheed-California Company) participated individually in this phase of the program by providing technical personnel, airfoil design concepts, and airfoil models. The overall objectives of the ATAT program were (1) to provide the industry participants with the opportunity to test and compare their advanced airfoils with the latest NASA airfoils at flight de-

sign Reynolds numbers in the same facility, (2) to provide industry with experience in cryogenic wind-tunnel model design, construction, and testing techniques, (3) to expand the high Reynolds number airfoil data base, and (4) to provide each participating company with the opportunity to evaluate their current level of airfoil technology. The industry participants were encouraged to explore innovative airfoil designs which may, for instance, be subject to strong Reynolds number effects and, therefore, may not represent an attempt to achieve an optimum level of performance. Consequently, caution should be exercised in drawing conclusions regarding overall levels of technology from direct comparisons of the results obtained on the various airfoils.

The pressure distributions presented in this report are from the first NASA/U.S. industry airfoil study conducted in the ATAT program. The industry participant for this study was the Boeing Commercial Airplane Company (Boeing). Consistent with the first two overall objectives of the ATAT program, this study was planned to test a 10-percent thick advanced-technology airfoil designed and fabricated by Boeing. The airfoil was designated BAC I for the purposes of this test. The aerodynamic coefficient data from this test have been published in reference 5. At the time of the test, the pressure coefficient data and model coordinates were proprietary and have only recently been made available by Boeing for general release. The tests were conducted in the Langley 0.3-m TCT with a two-dimensional, 20- by 60-cm (8- by 24-in.) test section installed. The operating envelope of this transonic cryogenic pressure tunnel is described in reference 27. Test temperature was varied from ambient to cryogenic temperature (about 100 K) at pressures ranging from about 1.2 to 6 atm (1 atm = 101.3 kPa). Mach number was varied from 0.40 to 0.80. These variables provided a Reynolds number range (based on airfoil model chord) from 4.4×10^6 to 50.0×10^6 . The pressure data are presented in plotted and tabulated formats. Also included in this report are the airfoil coordinates and comments on the model design and fabrication.

Symbols

The measurements and calculations were made in the U.S. Customary Units; the measurements are presented in the International System of Units (SI) with the U.S. Customary Units in parentheses. Factors relating these two systems of units can be found in reference 28. The symbols in parentheses are those used on computer-generated plots and tables in appendixes A through J.

AOA		angle of attack	y	(Y)	spanwise distance along model from centerline of tunnel and model, positive measured toward righthand side, cm (in.)
b	(B)	airfoil model span, cm (in.)			
c	(C)	airfoil model chord, cm (in.)			
	(CC)	section chord force coefficient from airfoil model pressures	α	(ALPHA)	uncorrected angle of attack, positive measured from tunnel centerline up to airfoil reference line, deg
c_d		section drag force coefficient from wake measurements			
	(CD1)	section drag coefficient from wake measurements for pitot tube at $\frac{y}{b/2} = 0.125$			
	(CD2)	section drag coefficient from wake measurements for pitot tube at $\frac{y}{b/2} = 0.0$			
	(CD3)	section drag coefficient from wake measurements for pitot tube at $\frac{y}{b/2} = -0.125$			
	(CD4)	section drag coefficient from wake measurements for pitot tube at $\frac{y}{b/2} = -0.375$			
	(CD5)	section drag coefficient from wake measurements for pitot tube at $\frac{y}{b/2} = -0.500$			
	(CDCOR1 through CDCOR5)	corrected values for CD1 through CD5			
c_m	(CM)	section pitching-moment coefficient about model quarter-chord point			
c_n	(CN)	section normal-force coefficient from model pressures			
	(C _p ,CP)	pressure coefficient			
M	(MACH)	free-stream Mach number			
	(MLOC)	local Mach number			
	(P,L)	local static pressure, kPa (psi)			
	(PT)	tunnel stagnation pressure, atm (1 atm = 101.3 kPa)			
R	(RC)	free-stream Reynolds number based on model chord			
	(TT)	tunnel stagnation temperature, K			
x	(X)	chordwise distance from leading edge of model, positive measured aft, cm (in.)			

Wind Tunnel and Model

Wind Tunnel

The tests were made in the 20- by 60-cm (8- by 24-in.) two-dimensional test section of the 0.3-m TCT. A photograph of the tunnel is shown in figure 1(a). A schematic drawing showing some physical characteristics of the tunnel is shown in figure 1(b). A photograph and sketch of the two-dimensional test section are shown in figure 2. In the photograph (fig. 2(a)), the plenum lid and test section ceiling have been removed to show model installation. The 0.3-m TCT is a continuous-flow, single-return, fan-driven transonic tunnel which uses nitrogen gas as the test medium. It is capable of operating at stagnation temperatures from about 80 K to about 327 K and stagnation pressures from slightly greater than 1 to 6 atm. Test-section Mach number can be varied from near 0 to 0.9. The ability to operate at cryogenic temperatures and a pressure of 6 atm provides an extremely high Reynolds number capability at relatively low model loadings.

The two-dimensional test section, which features a slotted floor and ceiling, contains computer-driven angle-of-attack and wake-survey-rake systems. The angle-of-attack system is capable of varying the angle of attack over a range of about 40°. The wake-survey rake, located just downstream of the model (fig. 2(a)), provides up to nine total-pressure measurements across half the width of the tunnel. These pressures are converted to drag levels and provide a convenient mechanism for determining the extent of two-dimensionality of the flow over the model. Additional design features and characteristics regarding the cryogenic concept in general and the 0.3-m TCT in particular are presented in references 29 and 30.

Model

The airfoil model used in this test is a 10-percent-thick, advanced-technology airfoil with a chord of 15.24 cm (6.0 in.). Table I presents the design coordinates for the airfoil. The model was designed and fabricated by Boeing in accordance with NASA structural and aerodynamic requirements for the ATAT

program models. The structural specifications included tolerance requirements for the model chord and span dimensions, a selection of material suitable for use at cryogenic temperatures, a safety factor of at least 3 at all operating conditions, Charpy impact strengths of at least 20.34 J (15.0 ft-lb) at 77 K, and compatibility with existing 0.3-m TCT sidewall turntables. The aerodynamic specifications required airfoil contour accuracies of ± 0.0025 cm (± 0.001 in.), surface finishes of $0.254 \mu\text{m}$ (0.00001 in.) or better, and a sufficient coverage of pressure orifices with diameters of about 0.025 cm (0.010 in.).

Model stress analysis. To meet the structural requirements, Boeing selected A-286 stainless steel for the model material. The Boeing stress analysis used a severe loading distribution anticipated at high angle of attack and a free-stream dynamic pressure of 196.31 kPa (4100 psf). Calculating stresses in various critical parts of the model with these loads and A-286 material properties by using classical methods gave safety factors of 8 or greater. A finite-element analysis of the model under load indicated a positive deflection of 0.0142 cm (0.006 in.) at the centerline section of the model. The decambering effect of trailing-edge movement under load was calculated to be only a 0.00097-cm (0.0004-in.) deflection with respect to the local airfoil chord; therefore, aeroelastic studies during the wind-tunnel test were considered unnecessary.

Model fabrication. The model was fabricated at the Boeing Aeronautical Laboratory model shop. Contouring was done in stages and the model was cryocycled (i.e., cooled to liquid-nitrogen temperature and warmed to ambient temperature) during the contouring phase to allow for material stabilization and reduce the possibility of model distortion during cryogenic testing. A "cover-plate" type of construction was used wherein trenches were cut into the upper and lower surfaces of the model block which had been machined to a slightly oversize contour for the aerodynamic surface. Holes were drilled in the bottom of these trenches to within approximately 0.127 cm (0.050 in.) of the opposite outside surface. Stainless-steel tubing was then soldered into all these holes with Eutectic EutecRod 157 solder. Except for the trailing-edge pressure orifice which required a final section of tubing with a 0.0254-cm (0.010-in.) outside diameter in order to remain within the cambered contour at the model trailing edge, tubing with a 0.0813-cm (0.032-in.) outside diameter and a 0.0406-cm (0.016-in.) inside diameter was used. The pressure tubes were then routed along the trenches and out a slot to the side of the model. Figure 3 is a photograph of the model during this phase

of construction. The cover plates were electron-beam welded over the trenches, and the model surfaces were machined to the final contour. Fifty-three static-pressure orifices, 0.0254 cm (0.010 in.) in diameter, were then cut into the model surface to meet the soldered tubes by using an electron discharge machine. The trailing-edge orifice used the inside diameter of the tube as the orifice. This technique for locating and cutting the orifices was made possible by using a computer-aided design system which improved the accuracy of the drawing and provided precise determination of the tangents to any point on the airfoil model surface. This then allowed the use of the leading edge of the model as a machining reference. Surface finishing was done by hand with fine-grit sandpaper.

Model accuracy. Experience has shown that many metals undergo drastic and irreversible changes in shape when exposed to a cryogenic environment. Therefore, it is standard practice that all airfoil models intended for testing in the 0.3-m TCT must be thermally cycled to cryogenic temperatures during the fabrication process so that they might stabilize before the final model contour validation and testing. Final contour and pressure orifice locations were checked by Boeing with a Brown & Sharpe Validator 200 probe. The actual airfoil contour (near the centerline) checked to within 0.00305 cm (0.0012 in.) and -0.00102 cm (-0.0004 in.) of the specified airfoil contour. These measurements were made at 10 chordwise locations on the upper and lower surfaces. The leading edge of the airfoil was checked with a template and the trailing-edge thickness was examined with a micrometer. The surface finish was measured by a surface roughness measuring device as $0.102 \mu\text{m}$ (0.000004 in.). Figure 4 is a schematic drawing which indicates the general locations of the orifices and the general shape of the airfoil section. The x/c and $y/b/2$ locations for each orifice are given in table II.

Just prior to installation in the tunnel, the model was cycled twice to cryogenic temperatures and back to ambient temperatures at a rate similar to actual operating conditions in the 0.3-m TCT. Visual and dye penetrant checks were made before and after the thermal cycling, and no flaws were found on the model. A photograph of the model installed in the test section of the 0.3-m TCT is shown in figure 5. (In this view, the plenum and test section ceiling have been removed and the model module is in the "raised" position above the test section.) The photograph shows the Boeing-selected transition tripping devices located at the 10-percent-chord line.

Test Apparatus and Procedures

Test Instrumentation and Apparatus

A detailed discussion of the instrumentation and procedures selected for the calibration and control of the 0.3-m TCT can be found in reference 31. Since, for airfoil model tests, the measured data are primarily (1) the pressure distributions around the airfoil model, (2) the definition of the wake defect, and (3) the angle of attack, the details of the relevant instrumentation are discussed herein.

Airfoil model pressures. The pressures on the airfoil model are measured with a scanning valve system capable of operating ten 48-port scanning valves. Because of the large changes in tunnel pressure over its operational range, commercially available, high-precision, variable-capacitance pressure transducers are used instead of conventional strain-gauge pressure transducers. The pressure transducers are located adjacent to the test section in order to reduce response time. To provide increased accuracy, the transducers are mounted on thermostatically controlled heater bases to maintain a constant temperature and on "shock" mounts to reduce possible vibration effects. The electrical outputs from the transducers are connected to individual signal conditioners located in the tunnel control room. The signal conditioners have autoranging capability and have seven ranges available. As a result of the autoranging capability, the analog electrical output to the data acquisition system is kept at a high level, even though the pressure transducer may be operating at the low end of its range. The maximum range of these differential transducers is about ± 6.8 atm, with an accuracy of ± 0.25 percent of the reading from -25 percent to $+100$ percent of full scale.

Wake pressures. A vertically traversing survey mechanism is located on the left sidewall of the two-dimensional test section downstream of the turntables (fig. 2). The purpose of this mechanism is to move a total-pressure probe rake through the airfoil wake to survey the total pressures within the wake. Details of this survey rake are shown in figure 6. The survey mechanism has a traversing range of 25.4 cm (10 in.). The rake support can be located with the measurement plane of the rake at either of two tunnel stations, 21.0 cm (8.3 in.) or 26.0 cm (10.2 in.). For this test, the wake survey measurements were made at the 26.0-cm (10.2-in.) station, which placed the measurement plane about 1.1 chord lengths downstream of the airfoil trailing edge. The survey mechanism is driven by an electric stepper motor and is designed to operate at speeds from about 0.25 to about

15 cm/sec (0.1 to 6 in/sec). The stroke (that portion of the total traversing range used in a given survey) and speed of the survey mechanism can be controlled from the operator's panel in the control room to suit the research requirements. The vertical position of the rake is recorded by using the output from a digital shaft encoder geared to the survey mechanism. The wake survey mechanism is synchronized with the scanning valves so that the rake is moved to a different vertical location each time the scanning valves are advanced to a new port. This movement continues until the scanning valves complete their stepping, at which time the rake continues to step at a predetermined rate through the remaining portion of the wake. Nine total-pressure probes are located on the survey rake. However, only five were used in this test because of blockage or leaks in the remaining four tubes. The five were located at the following spanwise stations: $\frac{y}{b/2} = 0.125, 0.0, -0.125, -0.375,$ and -0.500 . Nine tunnel sidewall static-pressure taps are also provided in the measurement plane of the rake. Data from these are averaged for use in the determination of the momentum loss and, therefore, airfoil drag coefficient based on the method outlined in reference 32. The more sensitive individual differential pressure transducers, with a maximum range of ± 1.36 atm and of the type described previously, are used on each tube on the survey rake and for each of the sidewall taps.

Angle of attack. The angle-of-attack mechanism has a traversing range of $\pm 20^\circ$, which can be offset from 0° in either direction at model installation. The mechanism is driven by an electric stepper motor, which is connected through a yoke to the perimeter of both turntables. This arrangement drives both ends of the model through the angle-of-attack range to eliminate possible model twisting. The angular position of the turntables, and therefore the angle of attack of the model, is recorded by using the output from a digital shaft encoder geared to one of the turntables.

Test Program

The test program (R as a function of M) used in this investigation is shown in figure 7. The selection of test conditions was made by Boeing in an effort to overlap some of their existing experimental and theoretical work. The extent of the effort to establish transition effects (fixed and free), Reynolds number effects, and Mach number effects can be seen in this figure.

Test Procedures

Delay times. After model installation and instrumentation checkout and calibration, it is necessary to establish the delay times required for the sampling of the airfoil pressures. Both experience and theoretical analysis have shown that the delay times are strongly dependent on the tubing diameters downstream of the model orifice, the pressure change from one orifice to another, and the magnitude of the pressure to be measured. As a result of these studies, the general recommendation was made to keep the inside diameter of the tubing within the model to greater than 0.076 cm (0.030 in.). This would result in normal delay times on the order of 1 to 2 sec/orifice. However, this model had tubing with inside diameters of about 0.051 cm (0.020 in.), which was expected to cause significant increases in delay times. By following normal procedures to determine delay times, predicted or preliminary pressure distributions for highly loaded model conditions were used to establish levels of individual orifice pressures and changes in level from adjoining orifices. These "known" pressures were applied to the airfoil statically and with tunnel flow, and the response of the pressure measuring system (orifice, tubing, and transducer) was determined by recording, on a strip chart, the time and pressure transient for the pressure to reach a settled pressure. For this test, 98 percent of the known level was selected as the settled pressure, and the resulting time was identified as the appropriate delay time. This procedure defined some delay times up to or in excess of 9.95 sec/port, which was the maximum capability of the controller. The remaining ports were also above normal in delay times, but could be grouped at 3 sec/port. A capability of the pressure-scanning-valve controller to vary delay times for groups of orifices provides near-minimum time consumed with near-maximum accuracy for each orifice. The groupings and delay times for the model orifices for this test are as shown in the following table:

Orifices	Approximate x/c	Delay time, sec
1	0	9.9
2 to 19	0.01 to 0.54	3.0
20 to 23	0.58 to 0.70	9.9
24 to 27	0.75 to 0.88	3.0
28	0.92	9.9
29	1.0	3.0

The other spanwise orifices had similar delay times based on their x/c location. The resulting

total time for the average data point to be taken approached 6 min.

Use of wake rake. To provide maximum definition of the model wake, the stroke of the rake (lower to upper limits) and the number of steps within the stroke were generally changed for each test condition such as angle of attack or Mach number. The range of values for these variables was determined from initial experimental runs. An example of this variation is shown in figure 8 for $M \approx 0.76$.

Transition. Transition strips located on both the upper and lower surfaces were used during a portion of the testing to evaluate their effect on the aerodynamic characteristics of the model. The trips were aluminum disks, 0.159 cm (0.063 in.) in diameter, 0.00254 cm (0.001 in.) thick, and spaced on 0.38-cm (0.15-in.) centers. The disks were glued along the 10-percent chord line with Loctite Depend two-part adhesive. The glue bond added an additional thickness of approximately 0.00508 cm (0.002 in.). Figure 7 shows the test conditions for fixed transition.

Data Reduction and Quality

Data Reduction

The test Mach number is based on an average of the Mach number distributions measured as a function of Reynolds number at several longitudinal stations during the calibration of the "empty" test section. Mach number is corrected for real-gas effects which are included in the data-reduction process through the use of the thermodynamic properties of nitrogen gas calculated from the Beattie-Bridgeman equation of state. This equation of state has been shown in reference 33 to give essentially the same thermodynamic properties and flow calculation results in the temperature-pressure regime of the 0.3-m TCT as are given by the more complicated Jacobsen equation of state. Detailed discussions of real-gas effects when testing in cryogenic nitrogen are contained in references 34 and 35.

The pressures on the surface of the airfoil were measured with pressure-scanning valves. The raw data were obtained by sampling 5 scans/port for the first portion of the test. This sampling was later changed to 3 scans/port in order to reduce the time required to record a data point.

Section normal-force and pitching-moment coefficients are calculated from numerical integrations (based on the trapezoidal method) of the local surface pressure coefficient measured at each orifice multiplied by an appropriate weighting factor (incremen-

tal area). Drag coefficient is obtained from the wake-survey pressures by computing an incremental or point drag coefficient by the method of reference 32 for each rake tube pressure at each rake position. These point drag coefficients are then numerically integrated across the model wake according to the trapezoidal method. Specifically, the point drag coefficients are compared one by one with a "threshold" value of drag coefficient, which accounts for a nonzero pressure decrement outside the model wake. This threshold is determined from several wake profiles early in the test as well as from past experience with similar tests. For this test, the threshold value was 0.0002. If, in the integrating process, the individual coefficient is greater than or equal to the threshold, the weighting factor (incremental area) is applied and the incremental drag is included in the running sum of the total drag. If the individual coefficient is less than the threshold, the weighting factor is set equal to zero and the incremental drag is not included in the running sum of the total drag. The results of this integration are total drag coefficients for each of the five rake pitot tubes. The data-reduction program then provides a correction which subtracts that summed portion of the individual incremental drag coefficients within the wake which is attributable to the threshold level. These corrected values of total drag coefficient are listed as CDCOR1 through CDCOR5 in the tabulated data in appendixes A through J.

After the test, it was noticed that for some data points, small portions of the airfoil wake were missed in the rake traverse. In a few other cases, the data system erroneously recorded zero values for certain portions of the wake profile. In each of these cases, the wake profile was extrapolated or interpolated manually as needed to complete the profile. The resulting addition to the drag coefficient was generally less than one count (0.0001). The results from the data-reduction process are presented in table III. Specific notation is made of those points which were adjusted by the extrapolation or interpolation process.

Data Quality

Mach number fluctuations. In all wind-tunnel testing, and especially in transonic testing, the steadiness of the tunnel flow conditions, such as Mach number, has direct bearing on the quality of the final aerodynamic data. In table III, values of Mach number and Reynolds number are shown as average values for the specific points. Because the delay times for some of the groupings of pressure orifices were very high, the variation in average values of Mach

number and Reynolds number does not represent an inability to set the precise tunnel test conditions in the short term but rather indicates a long-term drift in the test conditions during the extended time required for the acquisition of a single data point during these tests. In addition to the drift in test conditions due to the data-acquisition time, two other factors have been identified as causes of the undesired variations in Mach and Reynolds numbers. First, the manual control of the pressure and temperature control systems resulted in some fluctuation in the level of the Mach number. Second, the electrical drive system of the 0.3-m TCT has some inherent speed-control problems that feed directly into the tunnel flow through the fan drive. In all three areas, corrective measures have been identified and instituted.

Repeatability of data. Several examples illustrating the degree of repeatability for the normal-force, pitching-moment, and drag-force coefficients are shown in figures 9, 10, and 11. The repeatability shown in these figures is considered to be generally good, although there is some scatter in the data at the higher angles of attack.

Evaluation of hysteresis effects. An airfoil may exhibit substantially different aerodynamic characteristics at a given test condition, such as angle of attack, when the test condition is "approached" from different directions. A very brief attempt to develop hysteresis was made during this investigation, and the results obtained are reflected in table III and figure 12. The hysteresis data points were obtained by increasing the model angle of attack until substantial separation occurred and then decreasing the angle of attack to the desired test condition before taking data. The data indicate an absence of hysteresis over the operational range of the airfoil.

Concluding Remarks

A wind-tunnel investigation, which represents the first NASA/U.S. industry two-dimensional airfoil study to be completed in the Advanced Technology Airfoil Tests (ATAT) program, has been completed in the Langley 0.3-Meter Transonic Cryogenic Tunnel (0.3-m TCT). This investigation was designed to (1) test a Boeing advanced-technology airfoil from low to flight-equivalent Reynolds numbers, (2) provide Boeing with experience in cryogenic wind-tunnel model design, construction and testing techniques, and (3) demonstrate the suitability of the 0.3-m TCT as an airfoil test facility.

All the objectives of this investigation were met. The pressure data from this investigation are

presented without analysis in plotted and tabular formats for each Mach number and Reynolds number combination and are intended for use in conjunction with the aerodynamic coefficient data published in NASA Technical Memorandum 81922.

NASA Langley Research Center
Hampton, VA 23665-5225
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Appendixes

Because of the uncertainty in lift-induced interference effects and solid and wake blockage effects (particularly in the presence of local supercritical flow), no corrections for wall effects were applied to the basic experimental data at the time of the test. However, recent work has produced one method for assessment/correction of the wall boundary effects. Results for some tests conducted in the 0.3-m TCT are summarized in reference 23.

The pressure data from this investigation are presented without analysis or corrections in plotted and tabular formats in appendixes A through J to be used in conjunction with the aerodynamic coefficient data in reference 5. Each appendix contains data for a given Mach number through the Reynolds number range. For each combination of Mach number and Reynolds number, the data are plotted for each angle of attack with the associated tabulations immediately following. The pressure data from the upper surface of the airfoil are plotted as open symbols and the data from the lower surface are plotted as solid symbols. Since it is the intent of the authors to make these appendixes convenient for the user, similar angles of attack are always plotted at the same location on the page (e.g., $\alpha \approx 3$ is always at the page center). This arrangement should help the reader to follow a trend at a constant value of α , even if data were not taken at some angles of attack. The following table indicates the parameters plotted in each appendix:

Appendix	Mach number	Reynolds number $\times 10^{-6}$	Page
Free transition			
A	0.40	4.4 and 30.0	9
B	0.60	4.4, 7.7, 14.0, and 30.0	20
C	0.70	4.4, 7.7, 14.0, 30.0, and 45.0	41
D	0.74	4.4, 7.7, 14.0, 30.0, and 45.0	67
E	0.76	4.4, ^a 7.7, 14.0, 30.0, and 45.0	91
F	0.78	4.4, 7.7, 14.0, 30.0, and 45.0	128
G	0.80	4.4, 7.7, ^b 14.0, 30.0, 45.0, and 50.0	153
Fixed transition			
H	0.70	4.4, 7.7, and 14.0	179
I	0.76	4.4, ^b 7.7, 14.0, and 30.0	195
J	0.80	4.4, 7.7, and 14.0	219

^aConditions for "hysteresis data" runs.

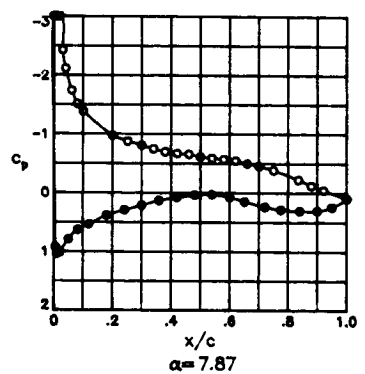
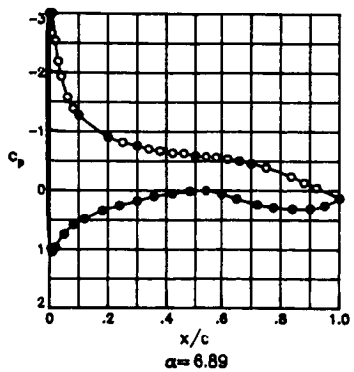
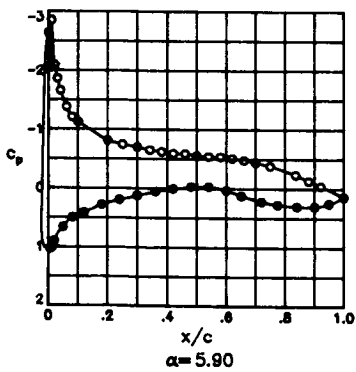
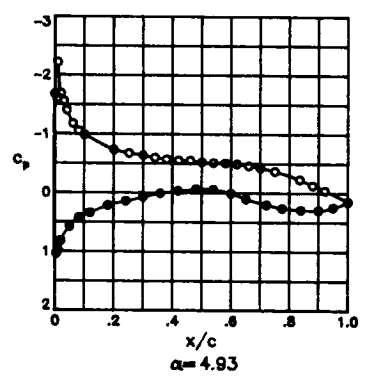
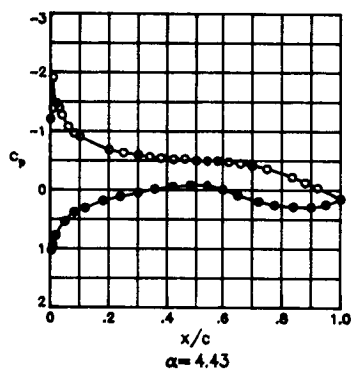
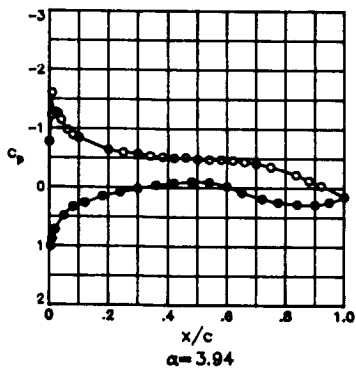
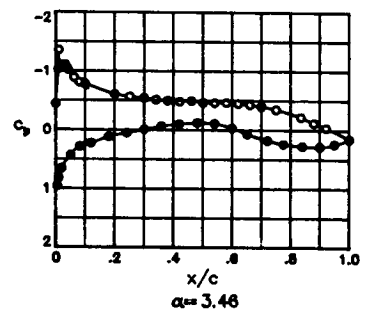
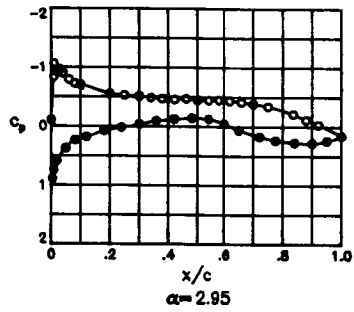
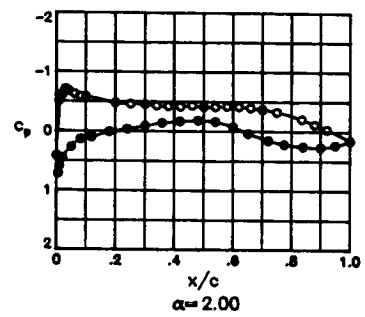
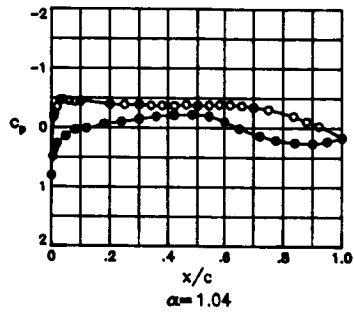
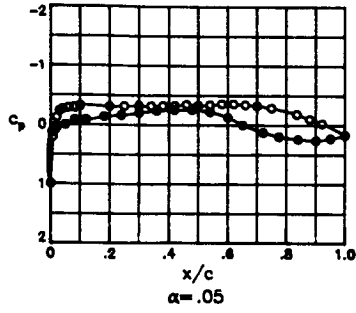
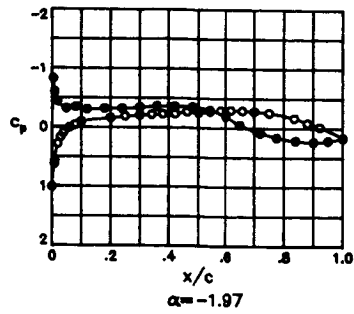
^bConditions for "repeat data" runs.

Appendix A

Pressure Data for $M = 0.40$; $R = 4.4 \times 10^6$ and 30.0×10^6 ; and Free Transition

The pressure measurements made on the Boeing BAC I airfoil are presented in coefficient form in graphs and tables in this appendix. The data are for a Mach number of 0.40; Reynolds numbers of 4.4×10^6 and 30.0×10^6 ; and free transition. The pressure data from the upper surface of the airfoil are plotted as open symbols, and the data from the lower surface are plotted as solid symbols.

TEST 122
 RUN 46
 MACH .401
 R 30.0×10^6



TEST RUN POINT	122 46 4	PT TT K	77.3843 102.1176 29.8630	PSI K MILLION	CN CM CC	.4347 -.0794 -.0081	CD1 CD2 CD3 CD4 CD5	.00651 .00663 .00662 .00634 .00632	CDCOR1 CDCOR2 CDCOR3 CDCOR4 CDCOR5	.00654 .00641 .00641 .00632 .00633		
		UPPER SURFACE		LOWER SURFACE		SPANWISE						
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/2	CP	P/L/PT	MLOC
0.0000	-.4176	.9376	.3064	0.0000	-.4176	.9376	.3064	.0500	-.3375	-.4997	.8475	.4947
.0083	-.5722	.8456	.4982	.0052	-.7253	.9687	.2148	.3957	-.3375	-.4124	.8561	.4791
.0097	-.6029	.8368	.5139	.0098	-.5951	.9556	.2570	.5008	-.3375	-.4151	.8554	.4804
.0203	-.8205	.8344	.5181	.0200	-.4489	.9409	.2978	.6048	-.3375	-.4107	.8557	.4797
.0390	-.7097	.8254	.5339	.0500	-.2679	.9230	.3421	.7003	-.3375	-.3716	.8592	.4733
.0400	-.6830	.8282	.5289	.0813	-.1390	.9100	.3715					
.0608	-.6227	.8340	.5188	.1199	-.1031	.9073	.3774					
.0800	-.5824	.8395	.5090	.1796	-.0130	.8979	.3974					
.1000	-.5752	.8395	.5091	.2337	-.0335	.8927	.4082					
.1997	-.4729	.8488	.4923	.2995	-.0896	.8871	.4195					
.2500	-.4568	.8507	.4890	.3538	-.1385	.8824	.4289					
.2994	-.4498	.8510	.4885	.4193	-.1662	.8793	.4350					
.3402	-.4277	.8537	.4831	.4793	-.1782	.8787	.4362					
.3795	-.4196	.8545	.4821	.5394	-.1570	.8806	.4324					
.4201	-.4126	.8552	.4806	.5994	-.0723	.8891	.4155					
.4598	-.4089	.8533	.4842	.6507	.0406	.9002	.3927					
.4996	-.4058	.8532	.4807	.7203	.1579	.9116	.3680					
.5397	-.4086	.8532	.4807	.7743	.2226	.9183	.3531					
.5795	-.4174	.8546	.4818	.8394	.2812	.9223	.3439					
.6197	-.4101	.8547	.4816	.8996	.2740	.9232	.3417					
.6598	-.3911	.8576	.4763	.9492	.2365	.9200	.3491					
.6997	-.3651	.8613	.4695	1.0000	.1615	.9126	.3559					
.7493	-.3227	.8637	.4650									
.8353	-.2027	.8766	.4404									
.8791	-.1102	.8855	.4228									
.9212	-.0265	.8931	.4074									
1.0000	.1615	.9126	.3659									

TEST RUN POINT	122 46 5	PT TT K	77.3845 102.0612 29.9380	PSI K MILLION	CN CM CC	.5308 -.0801 -.0178	CD1 CD2 CD3 CD4 CD5	.00666 .00676 .00650 .00643 .00620	CDCOR1 CDCOR2 CDCOR3 CDCOR4 CDCOR5	.00666 .00671 .00647 .00639 .00619		
		UPPER SURFACE		LOWER SURFACE		SPANWISE						
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/2	CP	P/L/PT	MLOC
0.0000	-.1056	.8857	.4222	0.0000	-.1056	.8857	.4222	.0500	-.3375	-.6221	.8346	.5177
.0083	-.8283	.8137	.5538	.0052	-.8956	.8856	.1451	.3957	-.3375	-.4562	.8509	.4886
.0097	-.10592	.7911	.5917	.0098	.7499	.9709	.2070	.5008	-.3375	-.4483	.8507	.4890
.0203	-.9138	.8044	.5696	.0200	.5940	.9552	.2579	.6048	-.3375	-.4342	.8522	.4862
.0390	-.9579	.7993	.5771	.0500	.3764	.9335	.3169	.7003	-.3375	-.3897	.8570	.4773
.0400	-.8961	.8061	.5667	.0813	.2352	.9192	.3510					
.0608	-.7256	.8183	.5486	.1199	.1845	.9143	.3620					
.0800	-.7222	.8239	.5370	.1796	.0793	.9036	.3853					
.1000	-.7007	.8256	.5333	.2397	.0213	.8992	.3948					
.1997	-.5499	.8417	.5051	.2995	-.0361	.8928	.4079					
.2500	-.5241	.8449	.4994	.3588	-.0924	.8877	.4183					
.2994	-.5054	.8468	.4960	.4193	-.1226	.8847	.4243					
.3402	-.4410	.8494	.4913	.4793	-.1418	.8829	.4278					
.3795	-.4657	.8498	.4906	.5394	-.1247	.8838	.4262					
.4201	-.4454	.8512	.4880	.5994	-.0468	.8918	.4100					
.4598	-.4658	.8507	.4889	.6507	.0619	.9030	.3866					
.4996	-.4408	.8523	.4849	.7203	.1728	.9138	.3631					
.5397	-.4408	.8531	.4845	.7743	.2359	.9202	.3496					
.5795	-.4437	.8534	.4840	.8394	.2728	.9242	.3394					
.6197	-.4341	.8538	.4832	.8996	.2827	.9249	.3377					
.6598	-.4122	.8556	.4800	.9492	.2412	.9205	.3470					
.6997	-.3824	.8579	.4758	1.0000	.1609	.9122	.3667					
.7493	-.3395	.8621	.4678									
.8353	-.2114	.8743	.4449									
.8791	-.1148	.8843	.4258									
.9212	-.0315	.8934	.4069									
1.0000	.1609	.9122	.3667									

TEST RUN POINT	122 46 6	PT TT K	77.3823 101.7475 30.6410	PSI K MILLION	CN CM CC	.5858 -.0804 -.0243	CD1 CD2 CD3 CD4 CD5	.00670 .00673 .00678 .00669 .00646	CDCOR1 CDCOR2 CDCOR3 CDCOR4 CDCOR5	.00683 .00676 .00681 .00671 .00652		
		UPPER SURFACE		LOWER SURFACE		SPANWISE						
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/2	CP	P/L/PT	MLOC
0.0000	-.4352	.8522	.4862	0.0000	-.4382	.8522	.4862	.0500	-.3375	-.7010	.8240	.5348
.0083	-1.0258	.7935	.5876	.0052	-.9638	.9920	.1078	.3957	-.3375	-.4813	.8468	.4960
.0097	-1.3318	.7609	.6408	.0098	.8208	.9780	.1796	.5008	-.3375	-.4722	.8482	.4934
.0203	-1.0937	.7866	.5991	.0200	.6599	.9621	.2369	.6048	-.3375	-.4526	.8503	.4898
.0390	-1.1070	.7862	.5999	.0500	.4331	.9393	.3020	.7003	-.3375	-.3974	.8563	.4788
.0400	-1.0229	.7960	.5870	.0813	.2853	.9250	.3374					
.0608	-.8546	.8393	.5620	.1199	.2293	.9192	.3511					
.0800	-.8042	.8482	.5497	.1796	.1193	.9081	.3759					
.1000	-.7706	.8493	.5445	.2397	.0561	.9013	.3933					
.1997	-.5976	.8357	.5158	.2995	-.0072	.8949	.4037					
.2500	-.5610	.8397	.5088	.3588	-.0657	.8892	.4153					
.2994	-.5385	.8428	.5033	.4193	-.0977	.8866	.4205					
.3402	-.5101	.8452	.4989	.4793	-.1198	.8842	.4255					
.3795	-.4941	.8474	.4950	.5394	-.1069	.8859	.4220					
.4201	-.4769	.8486	.4928	.5994	-.0300	.8932	.4073					
.4598	-.4491	.8481	.4937	.6507	.0736	.9040	.3847					
.4996	-.4609	.8515	.4875	.7203	.1818	.9150	.3604					
.5397	-.4583	.8513	.4878	.7743	.2436	.9210	.3469					
.5795	-.4413	.8529	.4898	.8394	.2782	.9243	.3392					
.6197	-.4468	.8511	.4882	.8996	.2881	.9247	.3382					
.6598	-.4255	.8538	.4832	.9492	.2445	.9206	.3477					
.6997	-.3951	.8566	.4782	1.0000	.1591	.9118	.3676					
.7493	-.3448	.8616	.4686									
.8353	-.2150	.8741	.4452									
.8791	-.1150	.8844	.4250									
.9212	-.0333	.8921	.4094									
1.0000	.1591	.9119	.3676									

Appendix B

Pressure Data for $M = 0.60$; $R = 4.4 \times 10^6$, 7.7×10^6 , 14.0×10^6 , and 30.0×10^6 ; and Free Transition

The pressure measurements made on the Boeing BAC I airfoil are presented in coefficient form in graphs and tables in this appendix. The data are for a Mach number of 0.60; Reynolds numbers of 4.4×10^6 , 7.7×10^6 , 14.0×10^6 , and 30.0×10^6 ; and free transition. The pressure data from the upper surface of the airfoil are plotted as open symbols, and the data from the lower surface are plotted as solid symbols.

TEST 122 PT 17.6862 PSI CM .8800
RUN 24 TT 173.6165 K CM -.0704
POINT 11 RC 4.4599 MILLION CC -.0608
MACH .6034
ALPHA 5.8994 DEG

CD1 .01452 CDCOR1 .01426
CD2 .01476 CDCOR2 .01445
CD3 .01481 CDCOR3 .01449
CD4 .01413 CDCOR4 .01388
CD5 .01354 CDCOR5 .01341

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC
0.0000	-.9001	.6054	.8778	0.0000	-.9001	.6034	.8778
.0083	-1.5838	.4701	1.0969	.0052	1.0786	.9968	.0669
.0097	-2.5042	.2829	1.4741	.0098	.9990	.9806	.1675
.0203	-2.5133	.2760	1.4763	.0200	.8634	.9539	.2605
.0300	-2.5092	.2809	1.4790	.0500	.6279	.9072	.3795
.0400	-2.4834	.2879	1.4617	.0813	.4871	.8785	.4340
.0608	-2.3328	.3148	1.3988	.1199	.3784	.8560	.4764
.0800	-2.1904	.3402	1.3433	.1796	.2581	.8325	.5184
.1000	-1.0403	.5726	.9292	.2397	.1657	.8153	.5480
.1997	-.8312	.6137	.8651	.2995	.0896	.7983	.5763
.2500	-.7833	.6247	.8481	.3588	.0151	.7843	.5994
.2994	-.7326	.6345	.8329	.4193	-.0515	.7707	.6213
.3402	-.6886	.6431	.8196	.4793	-.0741	.7661	.6287
.3795	-.6670	.6454	.8162	.5394	-.0711	.7653	.6300
.4201	-.6376	.6525	.8066	.5994	-.0027	.7818	.6034
.4598	-.6272	.6565	.7991	.6507	.1170	.8050	.5652
.4996	-.6008	.6627	.7896	.7203	.2268	.8274	.5272
.5397	-.5843	.6641	.7875	.7743	.2855	.8381	.5086
.5795	-.5655	.6691	.7798	.8394	.3070	.8430	.4998
.6197	-.5475	.6751	.7705	.8996	.2983	.8425	.5006
.6598	-.4938	.6816	.7606	.9492	.2452	.8297	.5232
.6997	-.4530	.6918	.7449	1.0000	.0708	.7975	.5776
.7493	-.3833	.7038	.7263				
.7933	-.2105	.7408	.6687				
.8391	-.1131	.7594	.6394				
.8912	-.0317	.7736	.6135				
1.0000	.0708	.7975	.5776				

SPANWISE				
X/C	Y/8/2	CP	P _L /PT	MLOC
.0500	-.3375	-1.9943	.3859	1.2504
.3957	-.3375	-.6492	.6543	.8026
.5008	-.3375	-.5986	.6642	.7873
.6048	-.3375	-.5343	.6721	.7752
.7003	-.3375	-.4519	.6935	.7422

TEST 122 PT 17.6868 PSI CM 1.0055
RUN 24 TT 172.9447 K CM -.0600
POINT 12 RC 4.4455 MILLION CC -.0761
MACH .6006
ALPHA 6.8966 DEG

CD1 .02350 CDCOR1 .02327
CD2 .02383 CDCOR2 .02352
CD3 .02355 CDCOR3 .02321
CD4 .02297 CDCOR4 .02276
CD5 .02248 CDCOR5 .02240

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC
0.0000	-1.2151	.5425	.9770	0.0000	-1.2151	.5425	.9770
.0043	-1.6810	.4501	1.1317	.0052	1.0780	.9970	.0656
.0097	-2.7432	.2419	1.5813	.0098	1.0391	.9894	.1233
.0203	-2.7946	.2339	1.6040	.0200	.9153	.9648	.2267
.0300	-2.7583	.2388	1.5901	.0500	.6868	.9201	.3468
.0400	-2.7315	.2474	1.5663	.0813	.5446	.8916	.4081
.0608	-2.6455	.2610	1.5297	.1199	.4341	.8701	.4502
.0800	-2.5428	.2831	1.4734	.1796	.3072	.8453	.4958
.1000	-2.1295	.3655	1.2909	.2397	.2092	.8267	.5284
.1997	-.8178	.6214	.8532	.2995	.1327	.8097	.5574
.2500	-.7886	.6275	.8437	.3588	.0574	.7949	.5819
.2994	-.7563	.6332	.8349	.4193	-.0043	.7823	.6026
.3402	-.7146	.6431	.8197	.4793	-.0410	.7761	.6126
.3795	-.6869	.6473	.8134	.5394	-.0449	.7744	.6153
.4201	-.6607	.6524	.8055	.5994	.0173	.7867	.5954
.4598	-.6470	.6545	.8022	.6507	.1262	.8079	.5603
.4996	-.6158	.6600	.7937	.7203	.2316	.8285	.5253
.5397	-.5976	.6655	.7854	.7743	.2880	.8407	.5040
.5795	-.5778	.6711	.7767	.8394	.3125	.8444	.4938
.6197	-.5302	.6758	.7695	.8996	.3011	.8417	.5022
.6598	-.5017	.6841	.7568	.9492	.2410	.8311	.5207
.6997	-.4524	.6936	.7422	1.0000	.0737	.7970	.5785
.7493	-.4061	.7066	.7221				
.7933	-.2149	.7405	.6693				
.8391	-.1198	.7595	.6392				
.8912	-.0431	.7736	.6167				
1.0000	.0737	.7970	.5785				

SPANWISE				
X/C	Y/8/2	CP	P _L /PT	MLOC
.0500	-.3375	-2.1807	.3493	1.3241
.3957	-.3375	-.6754	.6479	.8124
.5008	-.3375	-.6124	.6607	.7927
.6048	-.3375	-.5480	.6735	.7701
.7003	-.3375	-.4523	.6956	.7390

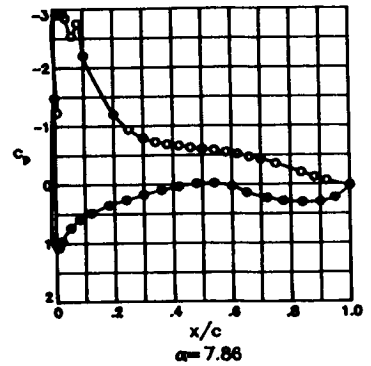
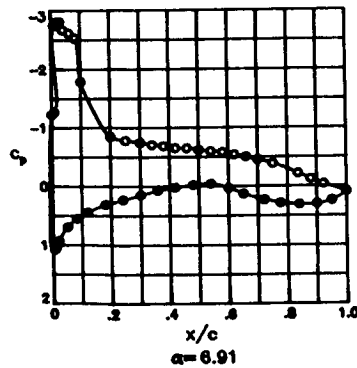
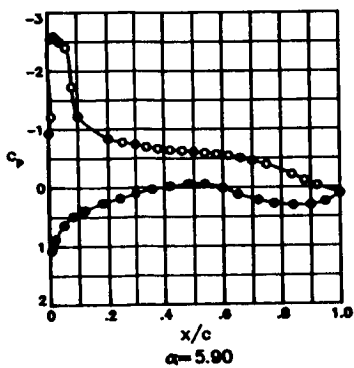
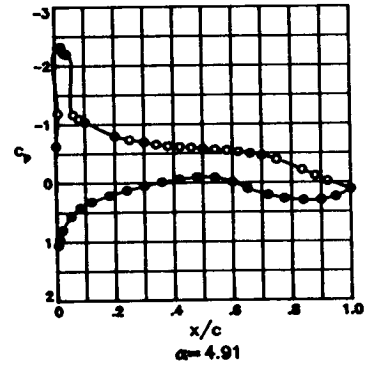
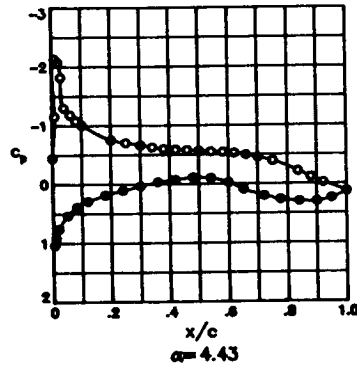
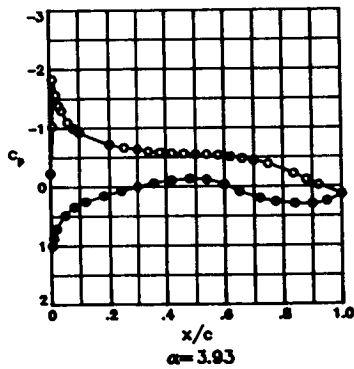
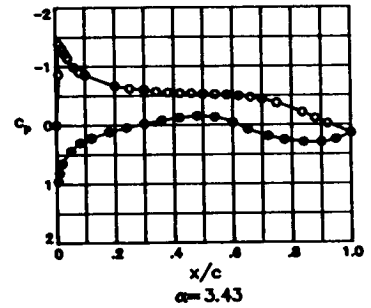
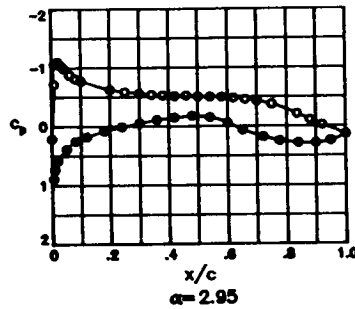
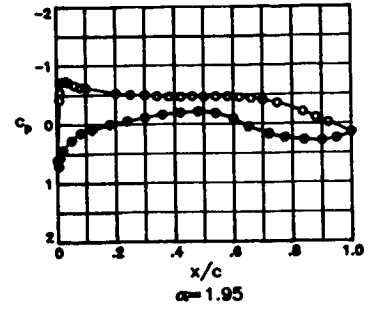
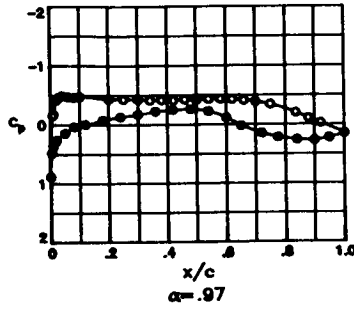
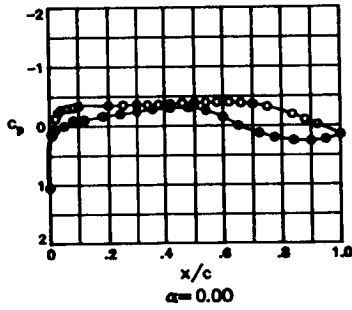
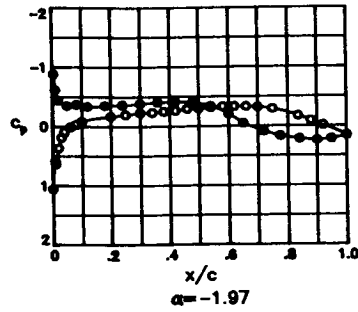
TEST 122 PT 17.6968 PSI CM 1.0611
RUN 24 TT 173.0249 K CM -.0511
POINT 13 RC 4.4428 MILLION CC -.0839
MACH .6008
ALPHA 7.8672 DEG

CD1 .03569 CDCOR1 .03531
CD2 .03647 CDCOR2 .03601
CD3 .03590 CDCOR3 .03501
CD4 .03548 CDCOR4 .03515
CD5 .03279 CDCOR5 .03263

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC
0.0000	-1.4493	.4962	1.0527	0.0000	-1.4493	.4962	1.0527
.0083	-1.7384	.4390	1.1515	.0052	1.0748	.9961	.0745
.0097	-2.0571	.2129	1.6671	.0098	1.0570	.9925	.1037
.0203	-2.8903	.2043	1.6946	.0200	.9514	.9713	.2045
.0300	-2.8496	.2098	1.6769	.0500	.7301	.9277	.3289
.0400	-2.8418	.2189	1.6500	.0813	.5896	.9001	.3907
.0608	-2.7767	.2325	1.6083	.1199	.4736	.8781	.4347
.0800	-2.7090	.2517	1.5545	.1796	.3432	.8497	.4878
.1000	-2.5585	.2697	1.5071	.2397	.2473	.8307	.5214
.1997	-.9199	.6017	.8637	.2995	.1595	.8152	.5480
.2500	-.8014	.6238	.8493	.3588	.0822	.7993	.5747
.2994	-.7527	.6324	.8362	.4193	.0236	.7869	.5951
.3402	-.7193	.6402	.8242	.4793	-.0250	.7780	.6096
.3795	-.6932	.6454	.8162	.5394	-.0327	.7765	.6121
.4201	-.6598	.6503	.8086	.5994	.0285	.7875	.5941
.4598	-.6394	.6542	.8027	.6507	.1287	.8074	.5613
.4996	-.6090	.6652	.7857	.7203	.2294	.8302	.5224
.5397	-.5873	.6678	.7819	.7743	.2857	.8404	.5046
.5795	-.5586	.6742	.7720	.8394	.3049	.8445	.4972
.6197	-.5253	.6810	.7615	.8996	.2929	.8423	.5011
.6598	-.4862	.6893	.7488	.9492	.2282	.8299	.5230
.6997	-.4371	.7000	.7322	1.0000	.0501	.7909	.5886
.7493	-.3673	.7103	.7162				
.7933	-.2082	.7415	.6677				
.8391	-.1258	.7580	.6416				
.8912	-.0576	.7728	.6186				
1.0000	.0501	.7909	.5886				

SPANWISE				
X/C	Y/8/2	CP	P _L /PT	MLOC
.0500	-.3375	-2.2992	.3294	1.3664
.3957	-.3375	-.6817	.6491	.8104
.5008	-.3375	-.6021	.6603	.7932
.6048	-.3375	-.5315	.6782	.7658
.7003	-.3375	-.4288	.7022	.7289

TEST 122
 RUN 32
 MACH .803
 R 7.7×10^6



ORIGINAL PAGE IS
OF POOR QUALITY

TEST 122 PT 17.6653 PSI CN .6798
 RUN 32 TT 117.1139 K CM -.0831
 POINT 7 RC 7.7792 MILLION CC -.0301
 MACH .6013
 ALPHA 3.9296 DEG

CD1 .00852 CDCOR1 .00843
 CD2 .00848 CDCOR2 .00837
 CD3 .00837 CDCOR3 .00825
 CD4 .00751 CDCOR4 .00743
 CD5 .00746 CDCOR5 .00743

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC
0.0000	-1.2279	.7393	.6717	0.0000	-1.2279	.7393	.6717
.0083	-1.0270	.5816	.9157	.0052	1.0009	.9817	.1630
.0097	-1.8250	.4221	1.1828	.0098	.8728	.9559	.2548
.0203	-1.5652	.4707	1.0966	.0200	.7098	.9239	.3384
.0300	-1.3768	.5102	1.0302	.0500	.4860	.8791	.4332
.0400	-1.2952	.5251	1.0058	.0813	.3429	.8503	.4873
.0608	-1.0960	.5636	.9440	.1199	.2530	.8327	.5185
.0800	-.9933	.5848	.9107	.1796	.1552	.8129	.5525
.1000	-.9261	.5975	.8909	.2397	.0731	.7974	.5784
.1997	-.7180	.6398	.8253	.2995	-.0035	.7832	.6016
.2500	-.6638	.6527	.8055	.3588	-.0583	.7724	.6191
.2994	-.6328	.6580	.7975	.4193	-.1028	.7630	.6343
.3402	-.5953	.6659	.7854	.4793	-.1310	.7577	.6427
.3795	-.5794	.6685	.7812	.5394	-.1176	.7600	.6389
.4201	-.5630	.6706	.7780	.5994	-.0286	.7688	.6119
.4598	-.5576	.6725	.7752	.6597	.0881	.8005	.5732
.4996	-.5429	.6744	.7723	.7203	.1997	.8220	.5368
.5397	-.5406	.6767	.7687	.7743	.2606	.8392	.5141
.5795	-.5286	.6793	.7640	.8394	.2946	.8423	.5015
.6197	-.5063	.6830	.7590	.8996	.2984	.8424	.5013
.6598	-.4753	.6883	.7509	.9492	.2508	.8325	.5188
.6997	-.4421	.6967	.7379	1.0000	.1339	.8103	.5569
.7493	-.3819	.7086	.7196				
.8353	-.2164	.7424	.6668				
.8791	-.1127	.7624	.6353				
.9212	-.0214	.7799	.6070				
1.0000	.1339	.8103	.5569				

SPANWISE			
X/C	Y/B/2	CP	P _L /PT
.0500	-1.3375	-1.1472	.5563
.3957	-1.3375	-.5680	.6714
.5008	-1.3375	-.5399	.6764
.6048	-1.3375	-.5092	.6836
.7003	-1.3375	-.4400	.6969

TEST 122 PT 17.6703 PSI CN .7324
 RUN 32 TT 117.4208 K CM -.0814
 POINT 8 RC 7.7320 MILLION CC -.0374
 MACH .5994
 ALPHA 4.4280 DEG

CD1 .00891 CDCOR1 .00882
 CD2 .00872 CDCOR2 .00861
 CD3 .00875 CDCOR3 .00862
 CD4 .00788 CDCOR4 .00782
 CD5 .00783 CDCOR5 .00781

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC
0.0000	-.4425	.6967	.7379	0.0000	-.4425	.6967	.7379
.0083	-1.1541	.5562	.9558	.0052	1.0367	.9890	.1257
.0097	-2.1317	.3670	1.2883	.0098	.9145	.9647	.2273
.0203	-2.0681	.3758	1.2709	.0200	.7573	.9340	.3141
.0300	-1.8154	.4276	1.1726	.0500	.5319	.8889	.4139
.0400	-1.2995	.5265	1.0035	.0813	.3875	.8606	.4683
.0608	-1.1808	.5509	.9442	.1199	.2908	.8429	.5006
.0800	-1.0772	.5747	.9265	.1796	.1856	.8206	.5394
.1000	-1.0052	.5852	.9100	.2397	.1069	.8040	.5673
.1997	-.7512	.6358	.8316	.2995	-.0346	.7909	.5890
.2500	-.7031	.6444	.8184	.3588	-.0345	.7767	.6122
.2994	-.6611	.6540	.8036	.4193	-.0745	.7697	.6235
.3402	-.6269	.6620	.7912	.4793	-.1112	.7634	.6335
.3795	-.6008	.6648	.7870	.5394	-.0982	.7642	.6323
.4201	-.5872	.6674	.7829	.5994	-.0203	.7796	.6075
.4598	-.5785	.6705	.7783	.6597	.0960	.8034	.5683
.4996	-.5599	.6752	.7710	.7203	.2051	.8255	.5308
.5397	-.5518	.6757	.7703	.7743	.2675	.8372	.5106
.5795	-.5418	.6787	.7657	.8394	.2976	.8437	.4991
.6197	-.5232	.6830	.7590	.8996	.3004	.8446	.4974
.6598	-.4938	.6913	.7463	.9492	.2429	.8347	.5149
.6997	-.4509	.6948	.7408	1.0000	.1285	.8101	.5572
.7493	-.3885	.7093	.7184				
.8353	-.2138	.7427	.6664				
.8791	-.1131	.7599	.6391				
.9212	-.0212	.7795	.6077				
1.0000	.1285	.8101	.5572				

SPANWISE			
X/C	Y/B/2	CP	P _L /PT
.0500	-1.3375	-1.1887	.5485
.3957	-1.3375	-.5937	.6668
.5008	-1.3375	-.5634	.6731
.6048	-1.3375	-.5296	.6803
.7003	-1.3375	-.4507	.6953

TEST 122 PT 17.6578 PSI CN .7828
 RUN 32 TT 117.4527 K CM -.0783
 POINT 9 RC 7.7289 MILLION CC -.0450
 MACH .6003
 ALPHA 4.9100 DEG

CD1 .00976 CDCOR1 .00963
 CD2 .00977 CDCOR2 .00960
 CD3 .00974 CDCOR3 .00954
 CD4 .00877 CDCOR4 .00866
 CD5 .00859 CDCOR5 .00854

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC
0.0000	-.6256	.6612	.7926	0.0000	-.6256	.6612	.7926
.0083	-1.1845	.5510	.9641	.0052	1.0570	.9930	.1007
.0097	-2.3019	.3319	1.3616	.0098	.9510	.9721	.2017
.0203	-2.3202	.3281	1.3699	.0200	.7998	.9420	.2935
.0300	-2.2117	.3473	1.3289	.0500	.5684	.8969	.3976
.0400	-2.1939	.3537	1.3155	.0813	.4205	.8680	.4545
.0608	-1.1616	.5573	.9540	.1199	.3201	.8475	.4922
.0800	-1.0877	.5699	.9341	.1796	.2135	.8267	.5289
.1000	-1.0380	.5801	.9181	.2397	.1278	.8091	.5588
.1997	-.7951	.6254	.8476	.2995	.0549	.7940	.5840
.2500	-.7340	.6405	.8243	.3588	-.0118	.7826	.6026
.2994	-.6929	.6488	.8116	.4193	-.0585	.7735	.6173
.3402	-.6526	.6560	.8005	.4793	-.0959	.7658	.6299
.3795	-.6258	.6597	.7948	.5394	-.0876	.7663	.6290
.4201	-.6099	.6631	.7897	.5994	-.0116	.7814	.6047
.4598	-.5978	.6662	.7849	.6597	.1032	.8045	.5665
.4996	-.5768	.6689	.7807	.7203	.2110	.8250	.5317
.5397	-.5652	.6723	.7754	.7743	.2697	.8373	.5104
.5795	-.5499	.6760	.7697	.8394	.3013	.8439	.4987
.6197	-.5249	.6787	.7657	.8996	.2974	.8419	.5023
.6598	-.4985	.6853	.7555	.9492	.2410	.8315	.5206
.6997	-.4626	.6947	.7411	1.0000	.1152	.8068	.5628
.7493	-.3904	.7681	.7203				
.8353	-.2145	.7407	.6696				
.8791	-.1154	.7614	.6367				
.9212	-.0225	.7774	.6111				
1.0000	.1152	.8068	.5628				

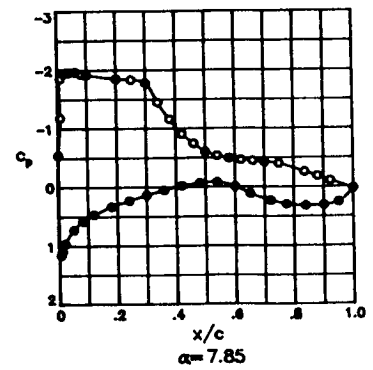
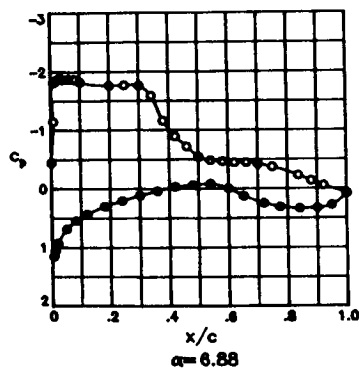
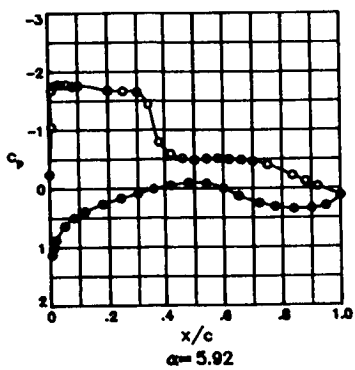
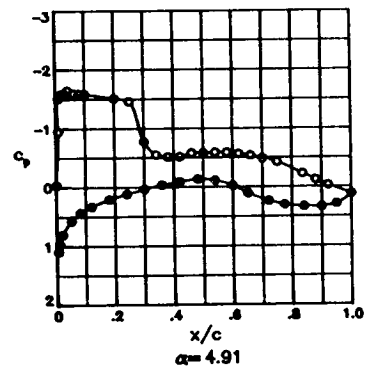
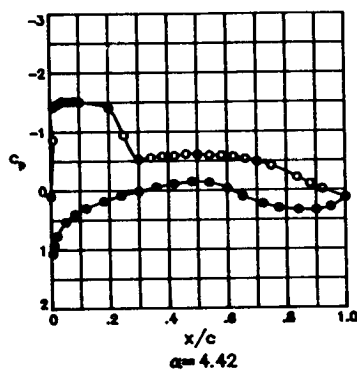
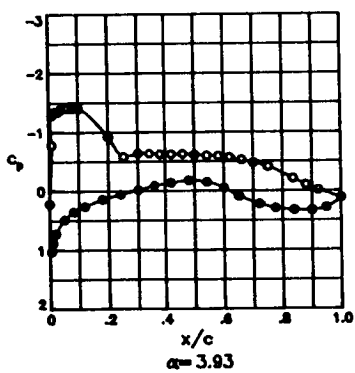
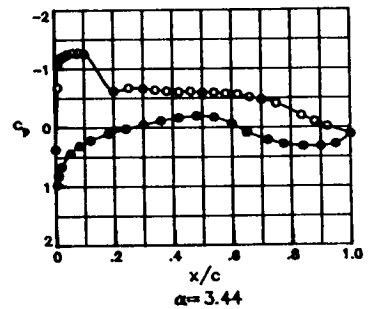
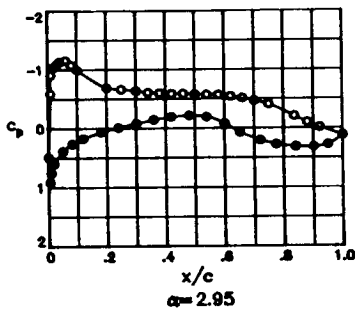
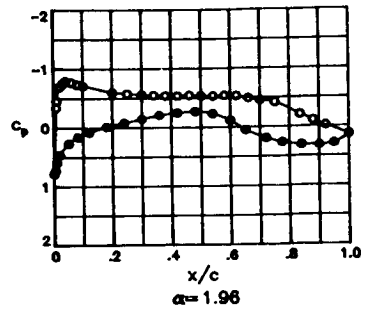
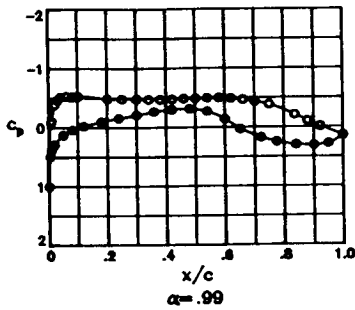
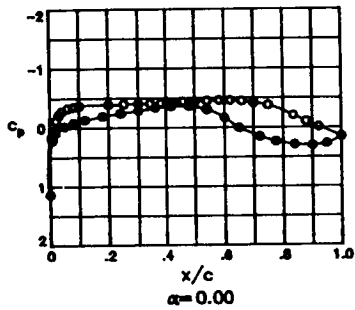
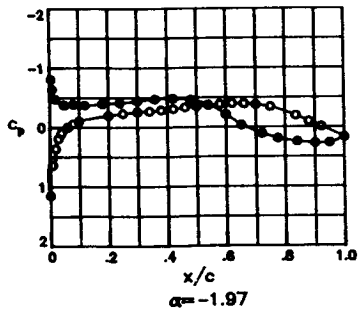
SPANWISE			
X/C	Y/B/2	CP	P _L /PT
.0500	-1.3375	-1.7684	.4318
.3957	-1.3375	-.6181	.6626
.5008	-1.3375	-.5858	.6703
.6048	-1.3375	-.5327	.6789
.7003	-1.3375	-.4561	.6922

Appendix C

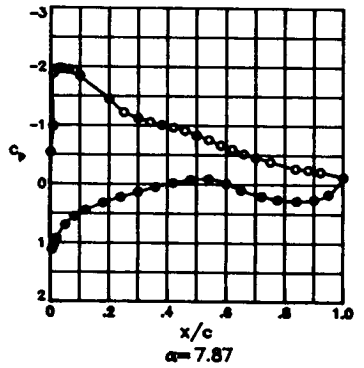
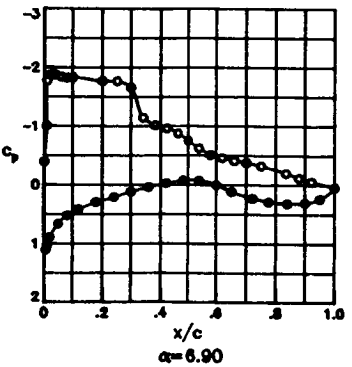
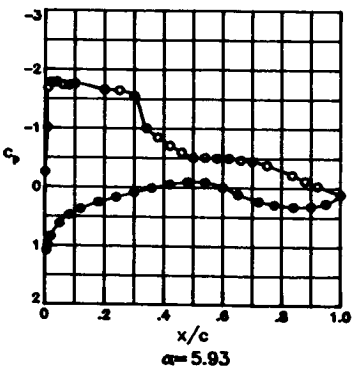
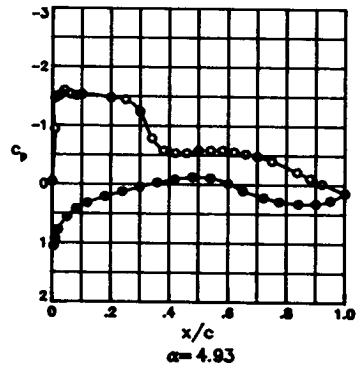
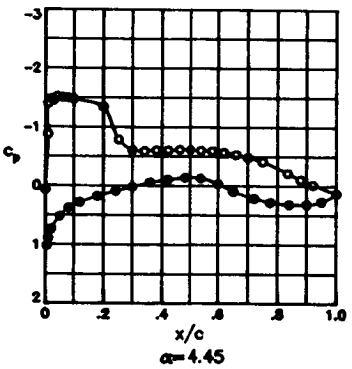
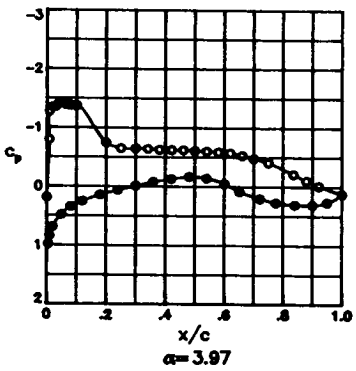
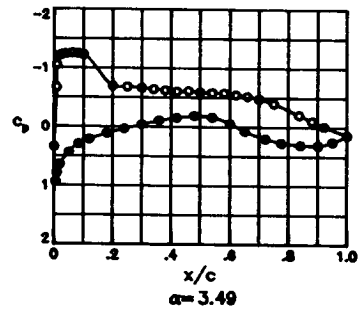
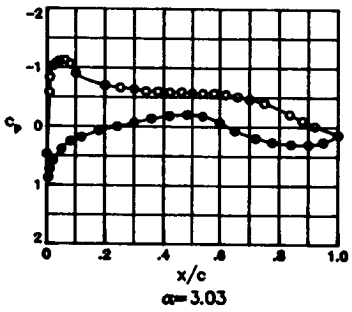
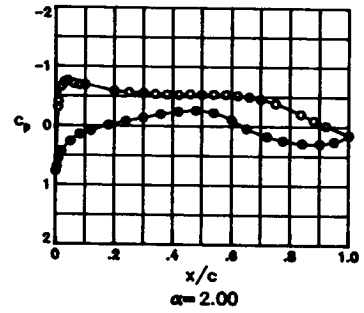
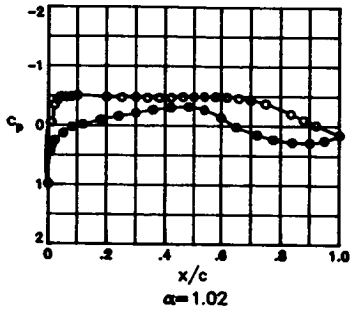
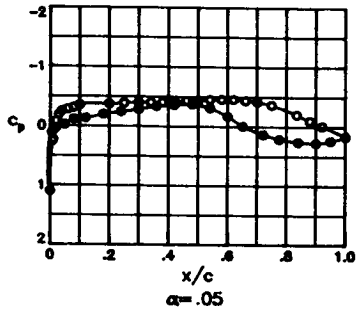
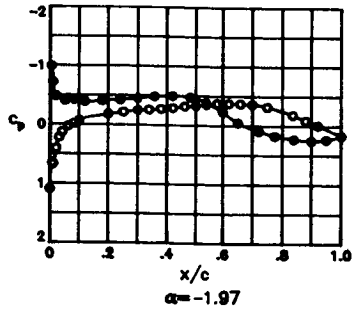
Pressure Data for $M = 0.70$; $R = 4.4 \times 10^6$, 7.7×10^6 , 14.0×10^6 , 30.0×10^6 , and 45.0×10^6 ; and Free Transition

The pressure measurements made on the Boeing BAC I airfoil are presented in coefficient form in graphs and tables in this appendix. The data are for a Mach number of 0.70; Reynolds numbers of 4.4×10^6 , 7.7×10^6 , 14.0×10^6 , 30.0×10^6 , and 45.0×10^6 ; and free transition. The pressure data from the upper surface of the airfoil are plotted as open symbols, and the data from the lower surface are plotted as solid symbols.

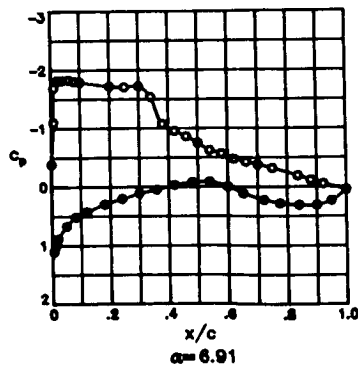
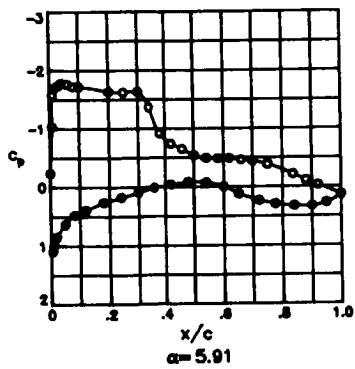
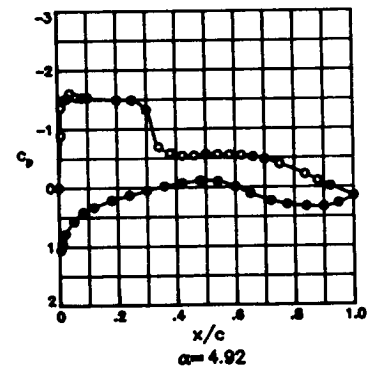
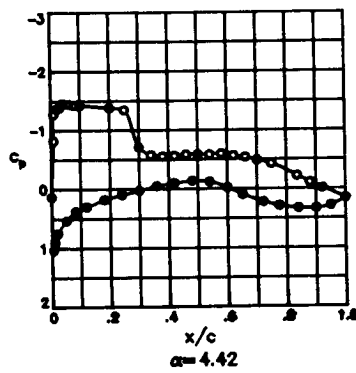
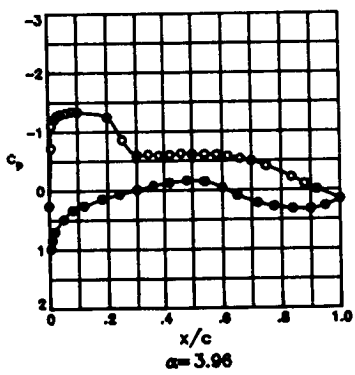
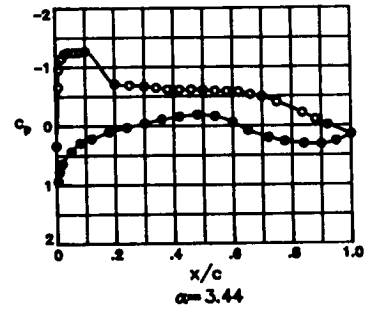
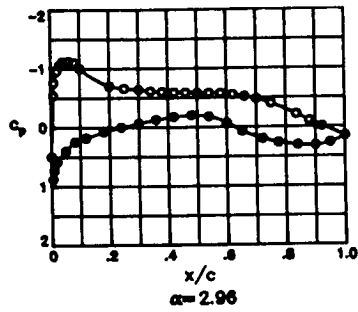
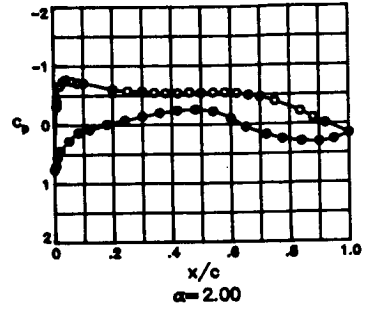
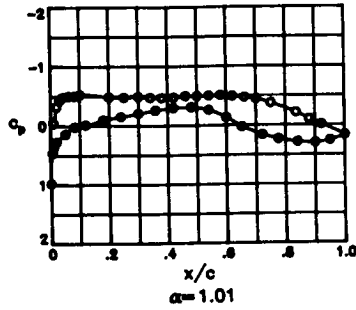
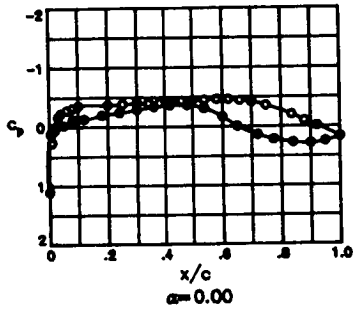
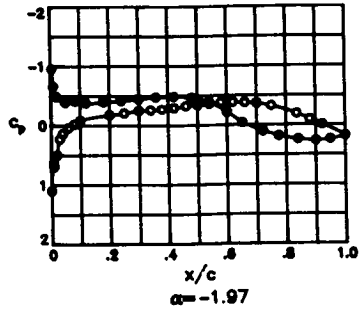
TEST 122
 RUN 23
 MACH .704
 R 4.4×10^6



TEST 122
 RUN 31
 MACH .704
 R 7.7×10^6



TEST 122
 RUN 38
 MACH .704
 R 14.0×10^6



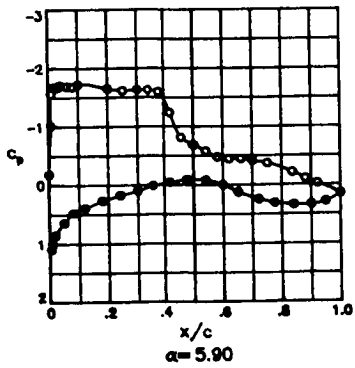
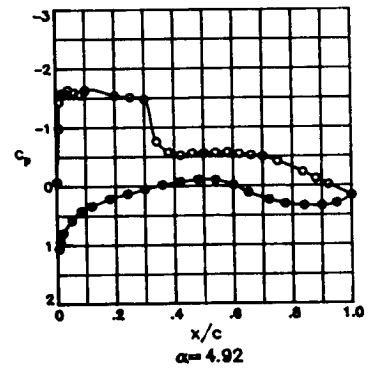
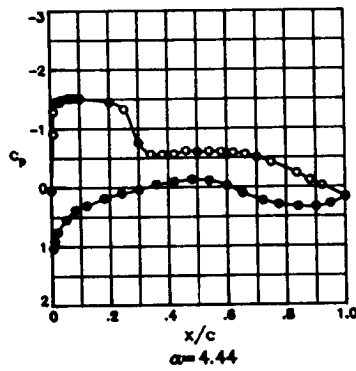
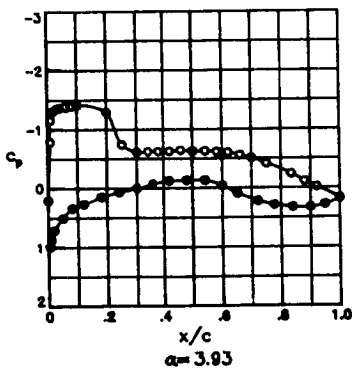
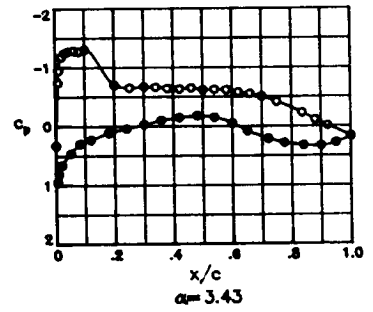
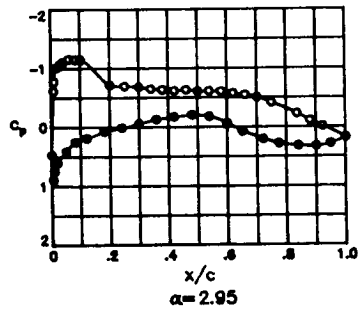
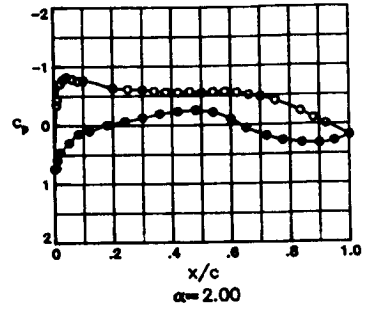
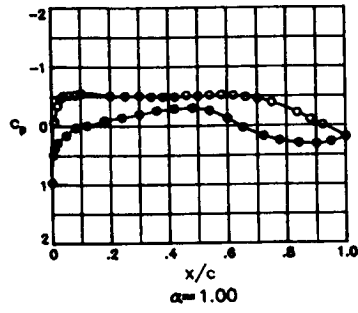
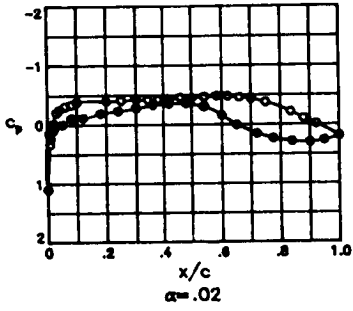
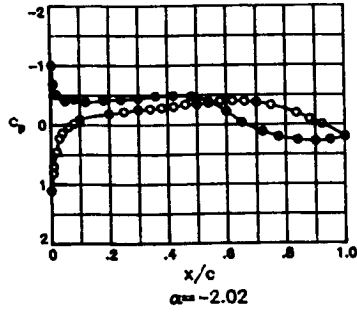
TEST	122	PT	23.908	PSI	CN	1.0411	CD1	.03418	CDCOR1	.03372
RUN	38	TT	101.4907	K	CM	-.0789	CD2	.03314	CDCOR2	.03260
POINT	11	PC	13.9930	MILLION	CC	-.0545	CD3	.03339	CDCOR3	.03289
		MACH	.7014				CD4	.03165	CDCOR4	.03133
		ALPHA	5.9085	DEG			CD5	.02662	CDCOR5	.02647

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/Z	CP	P/L/PT	MLOC
0.0000	-.2379	.6638	.7892	0.0000	-.2379	.6638	.7892	.6500	-.3375	-1.4585	-.3559	1.3117
.0093	-1.0310	.4687	1.1609	.0093	1.0950	.9915	.1107	.3957	-.3375	-.8571	-.5103	1.0308
.0097	-1.5789	.3293	1.3680	.0098	.9891	.9655	.2248	.5008	-.3375	-.5058	.5965	.8932
.0203	-1.7127	.2987	1.4371	.0200	.8466	.9305	.3230	.6048	-.3375	-.4733	.5999	.8879
.0300	-1.7394	.2931	1.4506	.0500	.6347	.8791	.4337	.7003	-.3375	-.4386	.6119	.8693
.0400	-1.7837	.2658	1.4680	.0813	.4751	.8401	.5060					
.0608	-1.7608	.2920	1.4530	.1199	.3905	.8193	.5421					
.0800	-1.7225	.3012	1.4314	.1796	.2614	.7848	.5997					
.1000	-1.7196	.2938	1.4489	.2397	.1756	.7636	.6339					
.1997	-1.6334	.3129	1.4044	.2995	.0920	.7419	.6683					
.2500	-1.6203	.3175	1.3941	.3588	.0141	.7231	.6977					
.2994	-1.6343	.3216	1.3848	.4193	-.0402	.7131	.7132					
.3432	-1.3582	.3865	1.2507	.4793	-.0827	.7012	.7318					
.3795	-.9123	.4964	1.0546	.5394	-.0756	.7028	.7292					
.4201	-.7353	.6434	.9771	.5994	.0006	.7238	.6967					
.4598	-.6485	.8578	.9541	.6507	.1228	.7492	.6567					
.4996	-.5389	.9855	.9104	.7203	.2377	.7784	.6101					
.5397	-.4420	1.0001	.8876	.7743	.2978	.7950	.5830					
.5795	-.4447	.9995	.8885	.8394	.3316	.8020	.5713					
.6197	-.4483	.8606	.8868	.8996	.3332	.8035	.5689					
.6598	-.4401	.6652	.8797	.9492	.2746	.7876	.5951					
.6997	-.4352	.6126	.8683	1.0000	.1411	.7529	.6510					
.7493	-.3789	.6291	.8427									
.8353	-.2699	.6691	.7811									
.8791	-.1056	.6929	.7444									
.9212	-.0220	.7124	.7144									
1.0000	.1411	.7529	.6510									

TEST	122	PT	22.6081	PSI	CN	1.1124	CD1	.05503	CDCOR1	.05467
RUN	38	TT	99.4443	K	CM	-.0781	CD2	.05338	CDCOR2	.05287
POINT	12	PC	14.0960	MILLION	CC	-.0587	CD3	.05911	CDCOR3	.05864
		MACH	.7042				CD4	.05172	CDCOR4	.05142
		ALPHA	6.9100	DEG			CD5	.04166	CDCOR5	.04152

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/Z	CP	P/L/PT	MLOC
0.0000	-.3425	.6291	.8443	0.0000	-.3825	.6281	.8443	.0500	-.3375	-1.5675	-.3326	1.3610
.0093	-1.1601	.4515	1.1309	.0092	1.1086	.9945	.0889	.3957	-.3375	-1.0157	.4646	1.1080
.0097	-1.6411	.2969	1.4414	.0098	1.0244	.9740	.1947	.5008	-.3375	-.7841	.9221	1.0115
.0203	-1.8143	.2711	1.5046	.0200	.8861	.9392	.3011	.6048	-.3375	-.4712	.6045	.8809
.0300	-1.8193	.2663	1.5170	.0500	.6737	.8866	.4189	.7003	-.3375	-.3747	.6257	.8479
.0400	-1.8214	.2667	1.5159	.0813	.5167	.8483	.4913					
.0608	-1.8290	.2677	1.5134	.1199	.4250	.8253	.5320					
.0800	-1.8063	.2720	1.5024	.1796	.2994	.7938	.5850					
.1000	-1.7878	.2755	1.4935	.2397	.2037	.7698	.6239					
.1997	-1.7176	.2968	1.4561	.2995	.1159	.7473	.6598					
.2500	-1.7644	.2922	1.4527	.3588	.0513	.7305	.6863					
.2994	-1.7201	.2924	1.4523	.4193	-.0267	.7128	.7138					
.3432	-1.5345	.3379	1.3494	.4793	-.0776	.7000	.7337					
.3795	-1.0720	.4436	1.1449	.5394	-.0630	.6931	.7442					
.4201	-.7490	.4834	1.0759	.5994	.0008	.7194	.7036					
.4598	-.8591	.5034	1.0414	.6507	.1173	.7474	.6598					
.4996	-.7463	.5331	.9937	.7203	.2268	.7752	.6153					
.5397	-.6225	.5662	.9408	.7743	.2879	.7916	.5886					
.5795	-.5648	.5753	.9265	.8394	.3136	.7952	.5827					
.6197	-.4741	.6005	.8871	.8996	.3120	.7962	.5810					
.6598	-.4226	.6147	.8649	.9492	.2325	.7773	.6119					
.6997	-.3753	.6269	.8462	1.0000	.0505	.7310	.6855					
.7493	-.3131	.6404	.8255									
.8353	-.1489	.6733	.7748									
.8791	-.1115	.6930	.7444									
.9212	-.0488	.7164	.7237									
1.0000	.0505	.7310	.6855									

TEST 122
 RUN 50
 MACH .704
 R 45.0×10^6



TEST 122 PT 78.4218 PSI CN 1.1116
 RUN 50 TT 164.9885 K CM -.0915
 POINT 10 RC 45.2350 MILLION CC -.0547
 MACH .7069
 ALPHA 5.9027 DEG

CD1 .03570 CDCDR1 .03237
 CD2 .03434 CDCDR2 .03169
 CD3 .03732 CDCDR3 .03466
 CD4 .02760 CDCDR4 .02565
 CD5 .02520 CDCDR5 .02349

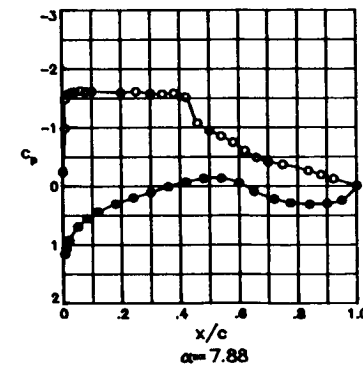
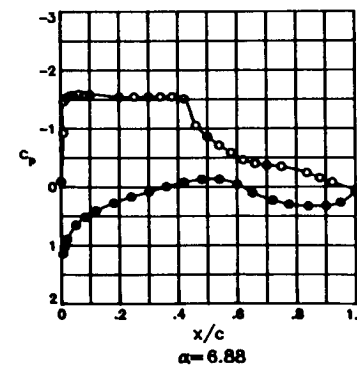
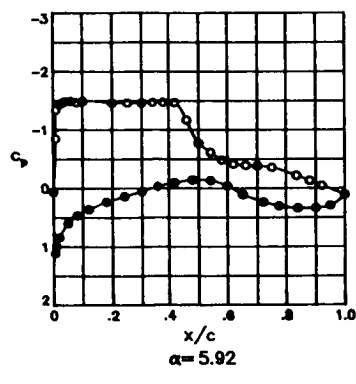
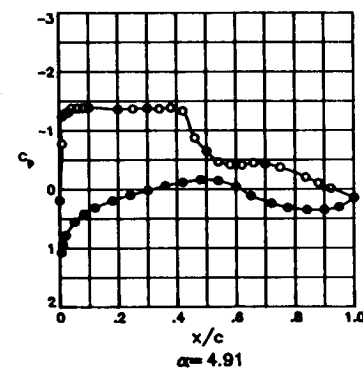
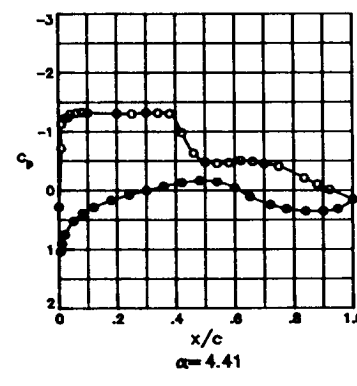
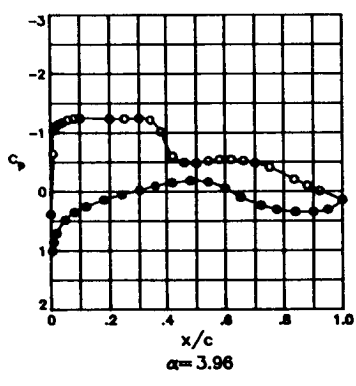
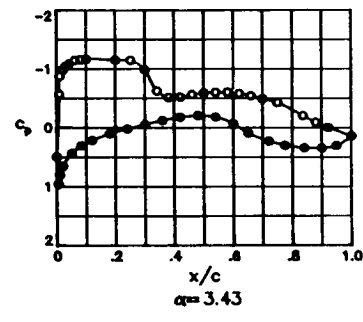
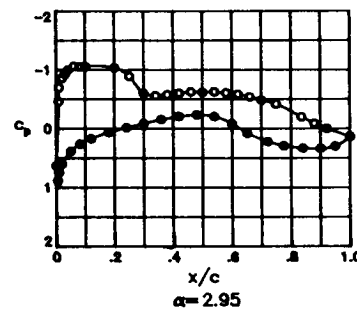
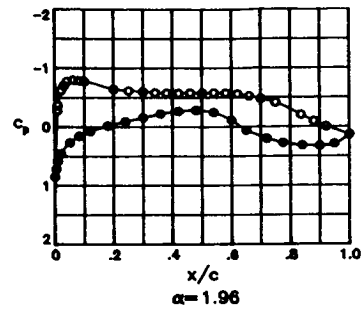
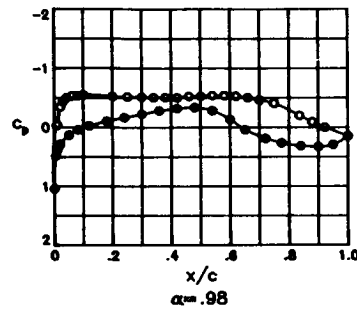
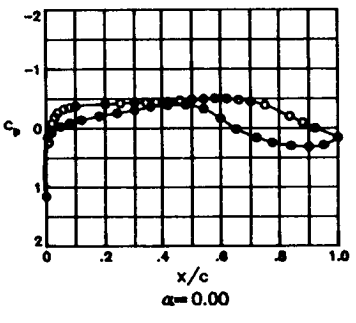
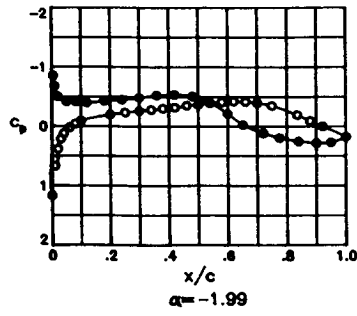
UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _z L/PT	MLOC	X/C	CP	P _z L/PT	MLOC	X/C	Y/B/2	CP	P _z L/PT	MLOC
0.0000	-.1822	.6714	.7803	0.0000	-.1822	.6714	.7803	.0500	-.3375	-1.3200	.3912	1.2452
.0083	-1.0176	.4626	1.1191	.0052	1.0996	.9923	.1055	.3957	-.3375	-1.5460	.3320	1.3658
.0097	-1.5618	.3330	1.3637	.0098	1.0027	.9682	.2164	.5008	-.3375	-.5931	.5723	.9343
.0203	-1.6813	.3622	1.4325	.0200	.8547	.9303	.3244	.6048	-.3375	-.4566	.6050	.8830
.0300	-1.6695	.2980	1.4424	.0500	.6441	.8783	.4367	.7003	-.3375	-.4343	.6145	.8681
.0400	-1.7113	.2966	1.4600	.0813	.4803	.8374	.5125					
.0608	-1.6932	.2951	1.4492	.1199	.3948	.8168	.5482					
.0800	-1.6787	.3011	1.4351	.1796	.2703	.7862	.5994					
.1000	-1.7202	.2917	1.4573	.2397	.1778	.7650	.6337					
.1997	-1.6464	.3115	1.4112	.2995	.1036	.7454	.6651					
.2500	-1.6127	.3133	1.4071	.3588	.0187	.7214	.7028					
.2994	-1.6280	.3126	1.4086	.4193	-.0404	.7081	.7236					
.3402	-1.6226	.3187	1.3949	.4793	-.0838	.6995	.7368					
.3795	-1.5903	.3189	1.3946	.5394	-.0757	.6978	.7395					
.4201	-1.2327	.4169	1.1968	.5994	.0073	.7229	.7004					
.4598	-.8057	.5164	1.0243	.6507	.1327	.7506	.6568					
.4994	-.6780	.5480	.9729	.7203	.2503	.7798	.6098					
.5397	-.5736	.5722	.9344	.7743	.3137	.7946	.5855					
.5795	-.4709	.6007	.8896	.8394	.3462	.8043	.5694					
.6197	-.4443	.6084	.8777	.8996	.3445	.8044	.5691					
.6598	-.4432	.6113	.8731	.9492	.2901	.7925	.5890					
.6997	-.4072	.6170	.8643	1.0000	.1559	.7556	.6488					
.7493	-.3644	.6281	.8471									
.8353	-.2100	.6673	.7866									
.8791	-.1003	.6933	.7464									
.9212	-.0165	.7173	.7092									
1.0000	.599	.7556	.6488									

Appendix D

Pressure Data for $M = 0.74$; $R = 4.4 \times 10^6$, 7.7×10^6 , 14.0×10^6 , 30.0×10^6 , and 45.0×10^6 ; and Free Transition

The pressure measurements made on the Boeing BAC I airfoil are presented in coefficient form in graphs and tables in this appendix. The data are for a Mach number of 0.74; Reynolds numbers of 4.4×10^6 , 7.7×10^6 , 14.0×10^6 , 30.0×10^6 , and 45.0×10^6 ; and free transition. The pressure data from the upper surface of the airfoil are plotted as open symbols, and the data from the lower surface are plotted as solid symbols.

TEST 122
 RUN 21
 MACH .745
 R 4.4×10^6



ORIGINAL PAGE IS
OF POOR QUALITY

TEST 122 PT 17.6913 PSI CN .0073
 RUN 21 TT 190.8895 K CM -.0946
 POINT 1 RC 4.4723 MILLION CC .0049
 MACH .7390
 ALPHA -1.9900 DEG

CD1 .00586 CDCOR1 .00575
 CD2 .00732 CDCOR2 .00721
 CD3 .00837 CDCOR3 .00825
 CD4 .00721 CDCOR4 .00711
 CD5 .00569 CDCOR5 .00564

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _s L/PT	MLOC	X/C	CP	P _s L/PT	MLOC
0.0000	1.1681	1.0062	0.0000	0.0000	1.1681	1.0062	0.0000
.0083	.6662	.8723	.4459	.0052	-.8579	.4645	1.1056
.0097	.6699	.8730	.4447	.0098	-.6824	.5125	1.0256
.0203	.3810	.7962	.5797	.0200	-.5056	.5586	.9512
.0300	.2110	.7502	.6539	.0500	-.4190	.5828	.9131
.0400	.1203	.7267	.6908	.0813	-.4171	.5842	.9109
.0608	.0166	.6997	.7327	.1199	-.3988	.5890	.9033
.0800	-.0434	.6837	.7573	.1796	-.4268	.5812	.9155
.1000	-.0913	.6707	.7773	.2397	-.4511	.5753	.9248
.1997	-.1976	.6415	.8221	.2995	-.4859	.5645	.9418
.2500	-.2345	.6310	.8382	.3588	-.5253	.5533	.9597
.2994	-.2655	.6228	.8509	.4193	-.5355	.5506	.9639
.3402	-.2832	.6198	.8555	.4793	-.5091	.5596	.9496
.3795	-.3042	.6147	.8634	.5394	-.3984	.5897	.9023
.4201	-.3221	.6082	.8735	.5994	-.2114	.6378	.8278
.4598	-.3527	.6006	.8853	.6507	-.0249	.6880	.7506
.4996	-.3687	.5967	.8914	.7203	.1133	.7252	.6932
.5397	-.3946	.5899	.9020	.7743	.1920	.7462	.6603
.5795	-.4131	.5842	.9108	.8394	.2524	.7619	.6355
.6197	-.4245	.5822	.9141	.8996	.2812	.7701	.6222
.6598	-.4179	.5844	.9105	.9492	.2675	.7668	.6276
.6997	-.3996	.5900	.9019	1.0000	.1732	.7424	.6662
.7493	-.3503	.6018	.8834				
.8353	-.1919	.6456	.8158				
.8791	-.0910	.6722	.7750				
.9212	-.0002	.6960	.7383				
1.0000	.1732	.7424	.6662				

SPANWISE				
X/C	Y/B/2	CP	P _s L/PT	MLOC
.0500	-.3375	.0248	.7021	.7209
.3957	-.3375	-.3160	.6125	.8669
.5008	-.3375	-.3656	.5992	.8875
.6048	-.3375	-.4130	.5858	.9084
.7003	-.3375	-.4009	.5907	.9007

TEST 122 PT 17.7289 PSI CN .2756
 RUN 21 TT 190.4784 K CM -.0996
 POINT 2 RC 4.4783 MILLION CC .0048
 MACH .7365
 ALPHA .0000 DEG

CD1 .00603 CDCOR1 .00590
 CD2 .00671 CDCOR2 .00658
 CD3 .00694 CDCOR3 .00682
 CD4 .00595 CDCOR4 .00585
 CD5 .00505 CDCOR5 .00499

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _s L/PT	MLOC	X/C	CP	P _s L/PT	MLOC
0.0000	1.1525	1.0023	0.0000	0.0000	1.1525	1.0023	0.0000
.0083	.2468	.7615	.6361	.0052	-.1512	.7360	.6762
.0097	.2343	.7581	.6414	.0098	-.1121	.7250	.6933
.0203	-.0852	.6725	.7745	.0200	.0723	.7141	.7104
.0300	-.1935	.6433	.8195	.0500	-.0299	.6882	.7503
.0400	-.2663	.6254	.8469	.0813	-.0940	.6706	.7774
.0608	-.3278	.6084	.8731	.1199	-.1421	.6580	.7967
.0800	-.3518	.6023	.8827	.1796	-.2062	.6407	.8234
.1000	-.3785	.5948	.8943	.2397	-.2557	.6279	.8430
.1997	-.4092	.5880	.9050	.2995	-.3055	.6155	.8622
.2500	-.4225	.5839	.9113	.3588	-.3603	.6005	.8855
.2994	-.4344	.5803	.9171	.4193	-.3906	.5919	.8989
.3402	-.4360	.5805	.9168	.4793	-.3965	.5909	.9004
.3795	-.4443	.5780	.9206	.5394	-.3269	.6092	.8720
.4201	-.4529	.5770	.9222	.5994	-.1673	.6527	.8049
.4598	-.4713	.5723	.9295	.6507	.0157	.7014	.7301
.4996	-.4832	.5690	.9348	.7203	.1607	.7396	.6706
.5397	-.4948	.5670	.9379	.7743	.2439	.7624	.6346
.5795	-.5041	.5631	.9442	.8394	.2956	.7752	.6141
.6197	-.4999	.5651	.9409	.8996	.3139	.7805	.6054
.6598	-.4791	.5704	.9326	.9492	.2845	.7726	.6182
.6997	-.4485	.5812	.9156	1.0000	.1545	.7396	.6706
.7493	-.3897	.5947	.8945				
.8353	-.2040	.6446	.8174				
.8791	-.0965	.6736	.7728				
.9212	-.0039	.6977	.7357				
1.0000	.1545	.7396	.6706				

SPANWISE				
X/C	Y/B/2	CP	P _s L/PT	MLOC
.0500	-.3375	-.2083	.6440	.8183
.3957	-.3375	-.4403	.5817	.9148
.5008	-.3375	-.4776	.5697	.9337
.6048	-.3375	-.4955	.5647	.9416
.7003	-.3375	-.4481	.5821	.9142

TEST 122 PT 17.7151 PSI CN .4018
 RUN 21 TT 190.2349 K CM -.1001
 POINT 3 RC 4.4830 MILLION CC .0000
 MACH .7365
 ALPHA .9800 DEG

CD1 .00743 CDCOR1 .00731
 CD2 .00667 CDCOR2 .00654
 CD3 .00671 CDCOR3 .00659
 CD4 .00621 CDCOR4 .00611
 CD5 .00526 CDCOR5 .00521

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _s L/PT	MLOC	X/C	CP	P _s L/PT	MLOC
0.0000	1.0392	.9721	.2015	0.0000	1.0392	.9721	.2015
.0083	-.0346	.6662	.7535	.0052	.4768	.8230	.5347
.0097	-.0493	.6834	.7578	.0098	.3775	.7966	.5792
.0203	-.3488	.6037	.8805	.0200	.2819	.7713	.6203
.0300	-.4399	.5797	.9179	.0500	.1298	.7303	.6852
.0400	-.5000	.5628	.9446	.0813	.0388	.7062	.7227
.0608	-.5364	.5532	.9598	.1199	-.0267	.6998	.7478
.0800	-.5356	.5549	.9570	.1796	-.1078	.6882	.7811
.1000	-.5523	.5503	.9644	.2397	-.1707	.6509	.8077
.1997	-.5259	.5562	.9550	.2995	-.2300	.6348	.8324
.2500	-.5190	.5593	.9502	.3588	-.2878	.6206	.8544
.2994	-.5183	.5601	.9489	.4193	-.3258	.6110	.8691
.3402	-.5102	.5618	.9461	.4793	-.3409	.6067	.8758
.3795	-.5496	.5622	.9455	.5394	-.2875	.6210	.8536
.4201	-.5480	.5624	.9452	.5994	-.1383	.6604	.7930
.4598	-.5277	.5577	.9526	.6507	.0371	.7072	.7210
.4996	-.5297	.5588	.9509	.7203	.1842	.7472	.6587
.5397	-.5406	.5539	.9587	.7743	.2620	.7666	.6279
.5795	-.5386	.5562	.9550	.8394	.3100	.7803	.6059
.6197	-.5303	.5576	.9528	.8996	.3230	.7832	.6010
.6598	-.5157	.5651	.9469	.9492	.2879	.7746	.6151
.6997	-.4647	.5749	.9254	1.0000	.1440	.7366	.6754
.7493	-.4497	.5910	.9103				
.8353	-.2033	.6429	.8206				
.8791	-.0948	.6714	.7762				
.9212	-.0037	.6966	.7374				
1.0000	.1440	.7366	.6754				

SPANWISE				
X/C	Y/B/2	CP	P _s L/PT	MLOC
.0500	-.3375	-.3799	.5973	.8904
.3957	-.3375	-.3055	.5635	.9434
.5008	-.3375	-.5274	.5579	.9524
.6048	-.3375	-.5338	.5564	.9547
.7003	-.3375	-.4638	.5742	.9265

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TEST 122 PT 17.6721 PSI CN .8282
 RUN 21 TT 190.6385 K CM -.0945
 POINT 7 RC 4.4438 MILLION CC -.0295
 MACH .7389
 ALPHA 3.9600 DEG

CD1 .01566 CDCOR1 .01490
 CD2 .01530 CDCOR2 .01468
 CD3 .01455 CDCOR3 .01359
 CD4 .01156 CDCOR4 .01090
 CD5 .00998 CDCOR5 .00957

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC
0.0000	-1.3851	.7980	.5768	0.0000	.3851	.7980	.5768
.0083	-1.6441	.5241	1.0067	.0052	1.0034	.9626	.2338
.0097	-1.0355	.4203	1.1854	.0098	.8618	.9255	.3342
.0203	-1.1149	.4609	1.2196	.0200	.7061	.8836	.4239
.0300	-1.1498	.3904	1.2417	.0500	.4805	.8222	.5361
.0400	-1.1739	.3794	1.2622	.0813	.3491	.7881	.5931
.0608	-1.2214	.3696	1.2824	.1199	.2531	.7623	.6547
.0800	-1.2391	.3644	1.2928	.1796	.1398	.7334	.6804
.1000	-1.2489	.3645	1.2927	.2397	.0522	.7097	.7172
.1997	-1.2371	.3691	1.2834	.2995	-.0237	.6907	.7465
.2500	-1.2382	.3661	1.2894	.3598	-.0975	.6696	.7789
.2994	-1.2455	.3657	1.2904	.4193	-.1519	.6561	.7998
.3402	-1.2155	.3703	1.2811	.4793	-.1854	.6451	.8165
.3795	-1.0170	.4229	1.1808	.5394	-.1632	.6508	.8078
.4201	-.4605	.5358	.9878	.5994	-.0558	.6807	.7619
.4598	-.4428	.5663	.9391	.6507	.0988	.7231	.6964
.4996	-.4813	.5687	.9352	.7203	.2344	.7587	.6103
.5397	-.5199	.5580	.9522	.7743	.3063	.7775	.5943
.5795	-.5412	.5518	.9621	.8394	.3446	.7874	.5943
.6197	-.5426	.5513	.9633	.8996	.3467	.7877	.5937
.6598	-.5226	.5575	.9529	.9492	.3071	.7778	.6098
.6997	-.4850	.5698	.9336	1.0000	.1434	.7331	.6808
.7493	-.4328	.5844	.9106				
.8353	-.2090	.6394	.8254				
.8791	-.1023	.6809	.7798				
.9212	-.0121	.6940	.7414				
1.0000	.1434	.7331	.6808				

SPANWISE			
X/C	Y/B/Z	CP	P/L/PT MLOC
.0500	-.3375	-.9182	.4518 1.1288
.3957	-.3375	-.8744	.4588 1.1164
.5008	-.3375	-.5114	.5601 .9488
.6048	-.3375	-.5509	.5492 .9661
.7003	-.3375	-.4789	.5689 .9350

TEST 122 PT 17.6898 PSI CN .8978
 RUN 21 TT 191.0984 K CM -.0958
 POINT 8 RC 4.4241 MILLION CC -.0344
 MACH .7370
 ALPHA 4.4093 DEG

CD1 .02067 CDCOR1 .02004
 CD2 .02063 CDCOR2 .02001
 CD3 .01919 CDCOR3 .01856
 CD4 .01482 CDCOR4 .01436
 CD5 .01295 CDCOR5 .01270

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC
0.0000	.2798	.7705	.6216	0.0000	.2798	.7705	.6216
.0083	-.7189	.5051	1.0378	.0052	1.0373	.9712	.2046
.0097	-1.1293	.3922	1.2381	.0098	.9045	.9371	.3059
.0203	-1.2227	.3749	1.2787	.0200	.7482	.8958	.3994
.0300	-1.2354	.3715	1.2787	.0500	.5208	.8504	.4521
.0400	-1.2491	.3561	1.3098	.0813	.3867	.8171	.5279
.0608	-1.3141	.3511	1.3203	.1199	.2861	.7918	.5673
.0800	-1.3286	.3457	1.3316	.1796	.1694	.7418	.6173
.1000	-1.3214	.3465	1.3299	.2397	.0780	.7158	.7077
.1997	-1.3058	.3513	1.3197	.2995	.0014	.6976	.7359
.2500	-1.3010	.3489	1.3249	.3598	-.0731	.6759	.7694
.2994	-1.3241	.3477	1.3272	.4193	-.1355	.6620	.7906
.3402	-1.3167	.3488	1.3251	.4793	-.1671	.6531	.8042
.3795	-1.3064	.3480	1.3268	.5394	-.1504	.6556	.8005
.4201	-.9826	.4376	1.1539	.5994	-.0442	.6859	.7539
.4598	-.6328	.5291	.9985	.6507	.1075	.7254	.6929
.4996	-.4828	.5675	.9372	.7203	.2408	.7599	.6043
.5397	-.4600	.5768	.9225	.7743	.3140	.7812	.6386
.5795	-.4719	.5728	.9288	.8394	.3519	.7908	.5888
.6197	-.5070	.5657	.9399	.8996	.3498	.7914	.5877
.6598	-.4912	.5683	.9358	.9492	.3111	.7804	.6057
.6997	-.4539	.5760	.9236	1.0000	.1514	.7349	.6780
.7493	-.4092	.5907	.9007				
.8353	-.2094	.6412	.8226				
.8791	-.1033	.6709	.7769				
.9212	-.0135	.6927	.7418				
1.0000	.1514	.7349	.6780				

SPANWISE			
X/C	Y/B/Z	CP	P/L/PT MLOC
.0500	-.3375	-.9877	.4438 1.1590
.3957	-.3375	-1.1372	.3969 1.2290
.5008	-.3375	-.4964	.5656 .9401
.6048	-.3375	-.5056	.5645 .9419
.7003	-.3375	-.4555	.5755 .9246

TEST 122 PT 17.6872 PSI CN .9522
 RUN 21 TT 191.2834 K CM -.0953
 POINT 9 RC 4.4189 MILLION CC -.0385
 MACH .7373
 ALPHA 4.9071 DEG

CD1 .02792 CDCOR1 .02699
 CD2 .02824 CDCOR2 .02735
 CD3 .02601 CDCOR3 .02510
 CD4 .01967 CDCOR4 .01898
 CD5 .01900 CDCOR5 .01757

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC
0.0000	.1858	.7466	.6597	0.0000	.1858	.7466	.6597
.0083	-.7782	.4913	1.0608	.0052	1.0699	.9809	.1666
.0097	-1.2420	.3695	1.2827	.0098	.9323	.9439	.2892
.0203	-1.2420	.3511	1.3203	.0200	.7801	.9035	.3832
.0300	-1.3099	.3489	1.3248	.0500	.5520	.8434	.4491
.0400	-1.3761	.3325	1.3596	.0813	.4144	.8065	.5220
.0608	-1.3789	.3316	1.3615	.1199	.3112	.7784	.6086
.0800	-1.3804	.3287	1.3679	.1796	.1926	.7483	.6569
.1000	-1.3900	.3292	1.3669	.2397	.0989	.7233	.6960
.1997	-1.3599	.3350	1.3542	.2995	.0190	.7013	.7302
.2500	-1.3690	.3342	1.3559	.3598	-.0570	.6819	.7600
.2994	-1.3780	.3317	1.3613	.4193	-.1182	.6657	.7850
.3402	-1.3679	.3320	1.3607	.4793	-.1620	.6527	.8049
.3795	-1.3664	.3319	1.3630	.5394	-.1432	.6598	.7940
.4201	-1.3305	.3433	1.3646	.5994	-.0417	.6854	.7547
.4598	-.8724	.4676	1.1112	.6507	.1121	.7277	.6892
.4996	-.6483	.5214	1.0111	.7203	.2387	.7582	.6414
.5397	-.4661	.5753	.9248	.7743	.3137	.7812	.6043
.5795	-.4174	.5866	.9072	.8394	.3488	.7896	.5907
.6197	-.4105	.5842	.9108	.8996	.3475	.7868	.5952
.6598	-.4464	.5815	.9151	.9492	.3004	.7782	.6091
.6997	-.4308	.5858	.9084	1.0000	.1406	.7348	.6781
.7493	-.3749	.5949	.8941				
.8353	-.2065	.6409	.8230				
.8791	-.1061	.6846	.7705				
.9212	-.0158	.6922	.7443				
1.0000	.1406	.7348	.6781				

SPANWISE			
X/C	Y/B/Z	CP	P/L/PT MLOC
.0500	-.3375	-1.0385	.4223 1.1816
.3957	-.3375	-1.2943	.3556 1.3108
.5008	-.3375	-.5933	.5379 .9843
.6048	-.3375	-.4447	.5814 .9153
.7003	-.3375	-.4164	.5847 .9101

TEST 122	PT	17.6461	PSI	CM	1.0053	CD1	.04755	CDCOR1	.04656
RUN 29	TT	128.7048	K	CM	-0.936	CD2	.04809	CDCOR2	.04705
POINT 11	RC	7.7909	MILLION	CC	-0.008	CD3	.04479	CDCOR3	.04378
	MACH	.7436				CD4	.03201	CDCOR4	.03115
	ALPHA	5.8974	DEG			CD5	.02817	CDCOR5	.02766

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC	X/C	Y/8/2	CP	P _L /PT	MLOC
0.0000	.0627	.7100	.7172	0.0000	.0627	.7100	.7172	.0500	-.3375	-1.3230	.3404	1.3431
.0083	-.8235	.4727	1.0930	.0052	1.0637	.9781	1.779	.3957	-.3375	-1.4198	.3150	1.3988
.0097	-1.3194	.3412	1.3415	.0098	.9418	.9451	.2853	.5008	-.3375	-.7336	.4943	1.0563
.0203	-1.4022	.3158	1.3969	.0200	.8005	.9078	.3744	.6048	-.3375	-.4635	.5689	.9354
.0300	-1.4270	.3123	1.4048	.0500	.5828	.8492	.4890	.7003	-.3375	-.3979	.5867	.9074
.0400	-1.4665	.3004	1.4324	.0813	.4413	.8119	.5539					
.0608	-1.4355	.3105	1.4489	.1199	.3379	.7823	.6029					
.0800	-1.4114	.3116	1.4066	.1796	.2209	.7517	.6521					
.1000	-1.4470	.3040	1.4240	.2397	.1381	.7309	.6848					
.1997	-1.4115	.3110	1.4078	.2995	.0514	.7049	.7250					
.2500	-1.4206	.3106	1.4088	.3588	-.0314	.6837	.7578					
.2994	-1.4492	.3096	1.4111	.4193	-.0862	.6724	.7751					
.3402	-1.4352	.3104	1.4092	.4793	-.1386	.6569	.7989					
.3795	-1.4654	.3097	1.4108	.5394	-.1270	.6639	.8882					
.4201	-1.3957	.3210	1.3853	.5994	-.0415	.6829	.7590					
.4598	-.8963	.4530	1.1271	.6597	.0907	.7174	.7057					
.4996	-.7954	.4799	1.0807	.7203	.2111	.7495	.6554					
.5397	-.6936	.5076	1.0341	.7743	.2757	.7671	.6274					
.5795	-.6166	.5251	1.0055	.8394	.3094	.7745	.6156					
.6197	-.5103	.5540	.9590	.8996	.3037	.7730	.6179					
.6598	-.4429	.5723	.9300	.9492	.2430	.7569	.6438					
.6997	-.3730	.5906	.9015	1.0000	.0779	.7149	.7096					
.7493	-.3312	.5989	.8884									
.8353	-.1971	.6405	.8242									
.8791	-.1143	.6619	.7912									
.9212	-.0392	.6822	.7601									
1.0000	.0779	.7149	.7096									

TEST 122	PT	17.5935	PSI	CM	.9950	CD1	.07938	CDCOR1	.07864
RUN 29	TT	128.8769	K	CM	-1.024	CD2	.08266	CDCOR2	.08168
POINT 12	RC	7.7335	MILLION	CC	-.0329	CD3	.06605	CDCOR3	.06516
	MACH	.7406				CD4	.06058	CDCOR4	.05988
	ALPHA	6.8883	DEG			CD5	.04901	CDCOR5	.04857

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC	X/C	Y/8/2	CP	P _L /PT	MLOC
0.0000	-.0706	.6755	.7705	0.0000	-.0706	.6755	.7705	.0500	-.3375	-1.3987	.3170	1.3942
.0083	-.8752	.4606	1.1138	.0052	1.0932	.9867	1.389	.3957	-.3375	-.8640	.4614	1.1124
.0097	-1.4578	.3097	1.4109	.0098	.9887	.9587	.2462	.5008	-.3375	-.7703	.4928	1.0580
.0203	-1.5341	.2882	1.4613	.0200	.8354	.9168	.3547	.6048	-.3375	-.6018	.5351	.9894
.0300	-1.5254	.2839	1.4719	.0500	.6216	.8595	.4704	.7003	-.3375	-.4715	.5670	.9384
.0400	-1.5455	.2786	1.4850	.0813	.4798	.8224	.5361					
.0608	-1.5381	.2837	1.4724	.1199	.3746	.7934	.5849					
.0800	-1.5138	.2875	1.4632	.1796	.2506	.7612	.6369					
.1000	-1.5475	.2812	1.4785	.2397	.1578	.7393	.6777					
.1997	-1.4816	.2938	1.4480	.2995	.0728	.7115	.7149					
.2500	-1.4285	.3123	1.4049	.3588	-.0194	.6888	.7499					
.2994	-1.1970	.3762	1.2697	.4193	-.0847	.6725	.7749					
.3402	-.9651	.4371	1.1553	.4793	-.1396	.6573	.7983					
.3795	-.8586	.4600	1.1149	.5394	-.1518	.6503	.8091					
.4201	-.8404	.4706	1.0965	.5994	-.0512	.6811	.7618					
.4598	-.7797	.4828	1.0758	.6597	.0666	.7100	.7171					
.4996	-.7582	.4935	1.0577	.7203	.1849	.7446	.6632					
.5397	-.6960	.5018	1.0438	.7743	.2403	.7548	.6471					
.5795	-.6594	.5167	1.0192	.8394	.2682	.7651	.6307					
.6197	-.6011	.5327	.9932	.8996	.2458	.7593	.6399					
.6598	-.5225	.5435	.9758	.9492	.1501	.7323	.6825					
.6997	-.5001	.5626	.9455	1.0000	-.1634	.6513	.8075					
.7493	-.4334	.5788	.9198									
.8353	-.3160	.6122	.8678									
.8791	-.3144	.6073	.8755									
.9212	-.2672	.6220	.8527									
1.0000	-.1634	.6513	.8075									

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TEST 122	PT	21.9828	PSI	CM	-.0112	CD1	.00733	CDCOR1	.00725
RUN 37	TT	100.1267	K	CM	-.0917	CD2	.00726	CDCOR2	.00718
POINT 1	RC	14.6790	MILLION	CC	.0043	CD3	.00720	CDCOR3	.00712
	MACH	.7417				CD4	.00720	CDCOR4	.00715
	ALPHA	-1.9800	DEG			CD5	.00705	CDCOR5	.00703
UPPER SURFACE									
X/C	CP	P _r L/PT	MLOC	X/C	CP	P _r L/PT	MLOC	SPANWISE	
0.0000	1.1162	.9922	.1059	0.0000	1.1162	.9922	.1059	X/C	Y/B/2
.0083	.6599	.8704	.4504	.0052	-1.0132	.4217	1.1843	CP	P _r L/PT
.0697	.7028	.8812	.4296	.0098	-.6412	.9218	1.0121	MLOC	
.0203	.5083	.8293	.5249	.0200	-.4499	.9606	.9496		
.0300	.2452	.7596	.6404	.0500	-.4143	.8832	.9140		
.0400	.1558	.7355	.6784	.0813	-.4413	.5765	.9245		
.0608	.0510	.7079	.7214	.1199	-.4087	.5833	.9138		
.0800	-.0102	.6900	.7489	.1796	-.4314	.5806	.9180		
.1000	-.0750	.6755	.7714	.2397	-.4536	.5720	.9316		
.1997	-.1850	.6457	.8172	.2995	-.4878	.5650	.9426		
.2500	-.2239	.6347	.8341	.3588	-.5280	.5535	.9608		
.2994	-.2530	.6269	.8461	.4193	-.5397	.5504	.9659		
.3402	-.2736	.6217	.8542	.4793	-.5111	.5583	.9532		
.3795	-.2919	.6184	.8623	.5394	-.4025	.5869	.9083		
.4201	-.3094	.6116	.8667	.5994	-.2146	.6369	.8307		
.4598	-.3490	.6032	.8828	.6507	-.0310	.6897	.7526		
.4996	-.3582	.5992	.8890	.7203	.1155	.7255	.6941		
.5397	-.3828	.5918	.9006	.7743	.1976	.7467	.6607		
.5795	-.4061	.5866	.9087	.8394	.2555	.7629	.6351		
.6197	-.4103	.5855	.9104	.8996	.2785	.7690	.6252		
.6598	-.4032	.5877	.9069	.9492	.2515	.7621	.6364		
.6997	-.3863	.5915	.9010	1.0000	.1918	.7458	.6621		
.7493	-.3274	.6065	.8776						
.8353	-.1852	.6451	.8181						
.8791	-.0810	.6732	.7749						
.9212	.0680	.6972	.7379						
1.0000	.1918	.7458	.6621						

TEST 122	PT	21.9896	PSI	CM	.2622	CD1	.00723	CDCOR1	.00714
RUN 37	TT	100.1394	K	CM	-.0958	CD2	.00716	CDCOR2	.00707
POINT 2	RC	14.0460	MILLION	CC	.0055	CD3	.00710	CDCOR3	.00702
	MACH	.7388				CD4	.00715	CDCOR4	.00710
	ALPHA	-.0058	DEG			CD5	.00698	CDCOR5	.00696
UPPER SURFACE									
X/C	CP	P _r L/PT	MLOC	X/C	CP	P _r L/PT	MLOC	SPANWISE	
0.0000	1.1169	.9929	.1013	0.0000	1.1169	.9929	.1013	X/C	Y/B/2
.0083	.2955	.7753	.6152	.0052	.1369	.7337	.6812	CP	P _r L/PT
.0697	.2764	.7708	.6227	.0098	.0868	.7207	.7015	MLOC	
.0203	.0617	.6592	.7564	.0200	.0613	.7117	.7155		
.0300	-.1603	.6527	.8063	.0500	-.0205	.6910	.7474		
.0400	-.2274	.6362	.8318	.0813	-.1147	.6650	.7875		
.0608	-.2901	.6183	.8593	.1199	-.1318	.6603	.7946		
.0800	-.3245	.6090	.8737	.1796	-.1998	.6434	.8206		
.1000	-.3654	.5995	.8885	.2397	-.2486	.6292	.8425		
.1997	-.3979	.5893	.9044	.2995	-.2985	.6158	.8632		
.2500	-.4135	.5840	.9115	.3588	-.3569	.5998	.8880		
.2994	-.4295	.5820	.9159	.4193	-.3838	.5941	.8969		
.3402	-.4289	.5816	.9165	.4793	-.3894	.5921	.9001		
.3795	-.4362	.5793	.9202	.5394	-.3187	.6106	.8714		
.4201	-.4403	.5703	.9206	.5994	-.1595	.6356	.8050		
.4598	-.4718	.5698	.9351	.6507	.0082	.6975	.7375		
.4996	-.4734	.5701	.9346	.7203	.1500	.7357	.6790		
.5397	-.4891	.5663	.9406	.7743	.2316	.7576	.6434		
.5795	-.4974	.5669	.9396	.8394	.2828	.7729	.6190		
.6197	-.4941	.5657	.9415	.8996	.2976	.7756	.6146		
.6598	-.4768	.5692	.9360	.9492	.2628	.7657	.6305		
.6997	-.4412	.5833	.9185	1.0000	.1809	.7438	.6653		
.7493	-.3674	.5965	.8932						
.8353	-.2018	.6432	.8210						
.8791	-.0910	.6720	.7767						
.9212	.0627	.6964	.7392						
1.0000	.1809	.7438	.6653						

TEST 122	PT	21.9904	PSI	CM	-.3950	CD1	.00741	CDCOR1	.00730
RUN 37	TT	100.2590	K	CM	-.0973	CD2	.00735	CDCOR2	.00724
POINT 3	RC	14.0720	MILLION	CC	.0006	CD3	.00728	CDCOR3	.00717
	MACH	.7433				CD4	.00730	CDCOR4	.00723
	ALPHA	.9800	DEG			CD5	.00710	CDCOR5	.00715
UPPER SURFACE									
X/C	CP	P _r L/PT	MLOC	X/C	CP	P _r L/PT	MLOC	SPANWISE	
0.0000	1.0120	.9643	.2287	0.0000	1.0120	.9643	.2287	X/C	Y/B/2
.0083	.0087	.6465	.7389	.0052	.4490	.8130	.5529	CP	P _r L/PT
.0697	.0266	.6998	.7339	.0098	.3930	.7875	.5993	MLOC	
.0203	-.2822	.6174	.8608	.0200	.2683	.7644	.6326		
.0300	-.4403	.5850	.9110	.0500	-.1329	.7279	.6903		
.0400	-.4461	.5671	.9392	.0813	.0165	.6975	.7375		
.0608	-.4481	.5589	.9523	.1199	-.0172	.6881	.7519		
.0800	-.5129	.5553	.9581	.1796	-.1008	.6668	.7846		
.1000	-.5443	.5483	.9692	.2397	-.1595	.6519	.8076		
.1997	-.5246	.5532	.9613	.2995	-.2203	.6346	.8342		
.2500	-.5212	.5545	.9592	.3588	-.2827	.6183	.8594		
.2994	-.5161	.5544	.9594	.4193	-.3178	.6075	.8760		
.3402	-.5161	.5569	.9555	.4793	-.3343	.6054	.8794		
.3795	-.5120	.5585	.9536	.5394	-.2784	.6207	.8556		
.4201	-.5132	.5585	.9563	.5994	-.1321	.6583	.7978		
.4598	-.5385	.5482	.9693	.6507	.0305	.7007	.7324		
.4996	-.5324	.5507	.9654	.7203	.1703	.7387	.6733		
.5397	-.5461	.5477	.9702	.7743	.2471	.7597	.6401		
.5795	-.5225	.5440	.9761	.8394	.2964	.7718	.6207		
.6197	-.5405	.5493	.9676	.8996	.3075	.7759	.6141		
.6598	-.5059	.5582	.9535	.9492	.2675	.7650	.6316		
.6997	-.4691	.5683	.9365	1.0000	.1746	.7421	.6681		
.7493	-.3854	.5896	.9039						
.8353	-.2052	.6387	.8280						
.8791	-.0910	.6705	.7790						
.9212	.0090	.6951	.7411						
1.0000	.1746	.7421	.6681						

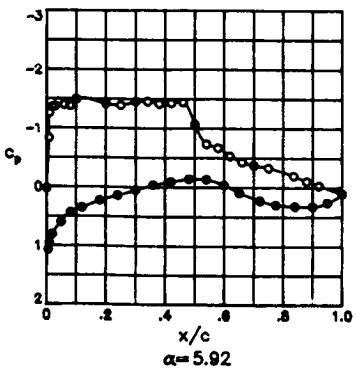
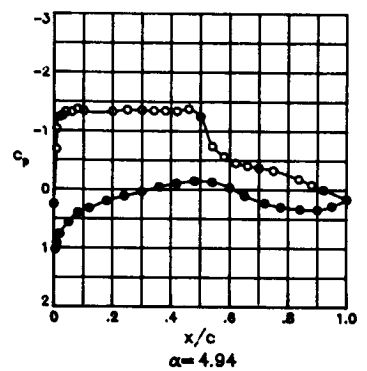
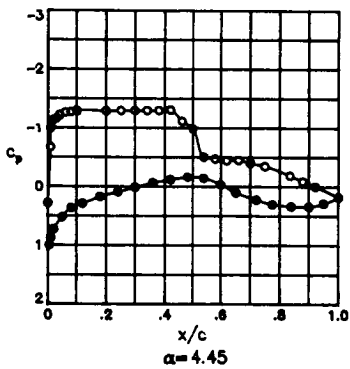
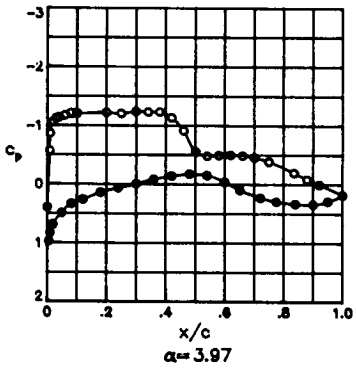
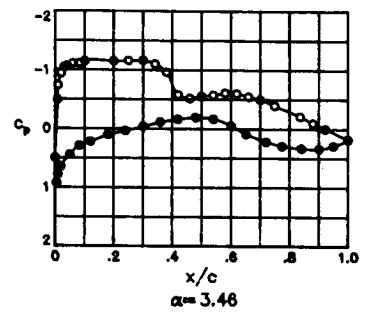
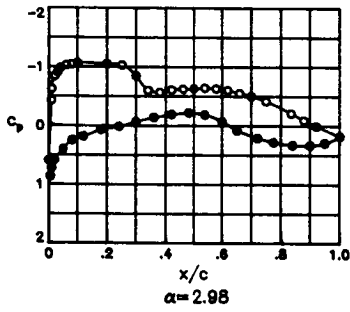
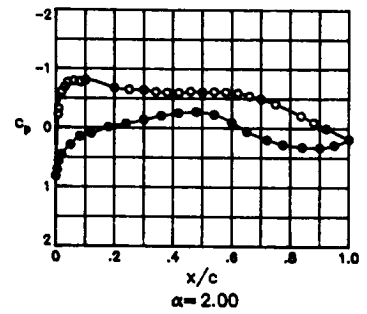
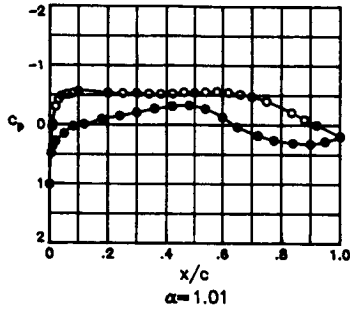
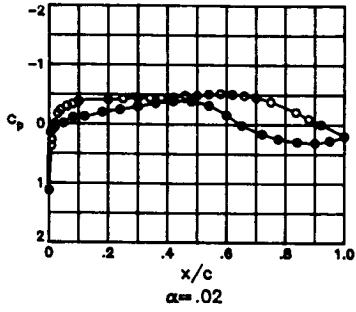
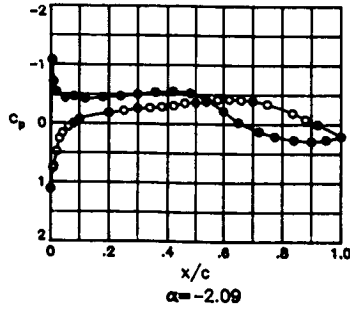
TEST 122	PT 22.8329	PSI	CM 1.0246	CD1 .04389	CDCOR1 .04344
RUN 37	TT 103.0308	K	CM -.0954	CD2 .04623	CDCOR2 .04569
POINT 10	RC 13.9400	MILLION	CC -.0417	CD3 .05660	CDCOR3 .05611
	MACH .7399			CD4 .03570	CDCOR4 .03543
	ALPHA 5.9170	DEG		CD5 .03410	CDCOR5 .03393

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/R/2	CP	P/L/PT	MLOC
0.0000	-.0618	.7121	.7147	0.0000	.0618	.7121	.7147	-.0500	-.3375	-1.2475	.3720	1.2792
.0083	-.8162	.4787	1.0837	.0052	1.0692	.9799	.1707	.3957	-.3375	-1.4503	.3139	1.4021
.0097	-1.2481	.3636	1.2960	.0098	.9470	.9463	.2823	.5008	-.3375	-.7637	.4986	1.0501
.0203	-1.3299	.3349	1.3557	.0200	.8032	.9090	.3723	.6048	-.3375	-.4961	.5647	.9431
.0300	-1.3920	.3244	1.3787	.0500	.5987	.8554	.4784	.7003	-.3375	-.3769	.5958	.8941
.0400	-1.4917	.3010	1.4318	.0813	.4390	.8112	.5558					
.0608	-1.4289	.3121	1.4061	.1199	.3533	.7884	.5937					
.0800	-1.4277	.3129	1.4045	.1796	.2325	.7583	.6422					
.1000	-1.4942	.3065	1.4330	.2397	.1409	.7324	.6831					
.1997	-1.4218	.3137	1.4026	.2995	.0514	.7075	.7219					
.2500	-1.4548	.3146	1.4006	.3588	-.0283	.6810	.7474					
.2994	-1.4446	.3118	1.4070	.4193	-.0905	.6717	.7770					
.3402	-1.4525	.3126	1.4050	.4793	-.1307	.6626	.7911					
.3795	-1.4688	.3114	1.4078	.5394	-.1272	.6651	.7871					
.4201	-1.4329	.3112	1.4082	.5994	-.0292	.6862	.7548					
.4598	-1.4805	.3790	1.2652	.6507	.0942	.7193	.7035					
.4996	-.8100	.4800	1.0815	.7203	.2179	.7534	.6500					
.5397	-.7206	.5030	1.0429	.7743	.2830	.7704	.6230					
.5795	-.5651	.5545	.9592	.8394	.3196	.7855	.5983					
.6197	-.5270	.5580	.9536	.8996	.3176	.7815	.6050					
.6598	-.4095	.5889	.9049	.9492	.2424	.7615	.6372					
.6997	-.3657	.6003	.8872	1.0000	.1000	.7233	.6974					
.7493	-.3121	.6097	.8725									
.8353	-.1872	.6455	.8173									
.8791	-.1035	.6696	.7804									
.9212	-.0370	.6820	.7613									
1.0000	.1000	.7233	.6974									

TEST 122	PT 22.8276	PSI	CM .9986	CD1 .07452	CDCOR1 .07391
RUN 37	TT 102.9082	K	CM -.1016	CD2 .07676	CDCOR2 .07602
POINT 11	RC 13.9900	MILLION	CC -.0328	CD3 .06207	CDCOR3 .06138
	MACH .7427			CD4 .06749	CDCOR4 .06704
	ALPHA 6.8947	DEG		CD5 .05707	CDCOR5 .05676

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/R/2	CP	P/L/PT	MLOC
0.0000	-.0413	.6824	.7606	0.0000	-.0413	.6824	.7606	-.0500	-.3375	-1.2538	.3585	1.3063
.0083	-.4858	.4639	1.1092	.0052	1.0875	.9846	.1494	.3957	-.3375	-.9051	.4557	1.1233
.0097	-1.3652	.3304	1.3656	.0098	.9776	.9543	.2598	.5008	-.3375	-.7269	.4914	1.0623
.0203	-1.4368	.3044	1.4239	.0200	.8430	.9196	.3486	.6048	-.3375	-.6135	.5325	.9945
.0300	-1.5063	.2942	1.4478	.0500	.6270	.8607	.4687	.7003	-.3375	-.4896	.5622	.9470
.0400	-1.5184	.2855	1.4688	.0813	.4718	.8200	.5409					
.0608	-1.5170	.2887	1.4609	.1199	.3941	.8035	.5687					
.0800	-1.5736	.2866	1.4661	.1796	.2447	.7568	.6447					
.1000	-1.5099	.2842	1.4714	.2397	.1543	.7333	.6816					
.1997	-1.4674	.2940	1.4483	.2995	.0617	.7067	.7231					
.2500	-1.4412	.3039	1.4250	.3588	-.0248	.6848	.7569					
.2994	-1.3193	.3475	1.3291	.4193	-.0900	.6732	.7748					
.3402	-1.3625	.4104	1.2051	.4793	-.1452	.6553	.8022					
.3795	-.9420	.4503	1.1328	.5394	-.1377	.6624	.7914					
.4201	-.8944	.4724	1.0945	.5994	-.0612	.6819	.7614					
.4598	-.7433	.4834	1.0749	.6507	.0674	.7131	.7132					
.4996	-.7482	.4956	1.0552	.7203	.1838	.7439	.6651					
.5397	-.6966	.5070	1.0362	.7743	.2492	.7601	.6395					
.5795	-.6480	.5213	1.0128	.8394	.2658	.7652	.6313					
.6197	-.5932	.5363	.9883	.8996	.2497	.7611	.6378					
.6598	-.5409	.5473	.9707	.9492	.1429	.7307	.6858					
.6997	-.4956	.5628	.9461	1.0000	-.1581	.6517	.8077					
.7493	-.4473	.5796	.9194									
.8353	-.3382	.6003	.8872									
.8791	-.2931	.6157	.8634									
.9212	-.2516	.6254	.8483									
1.0000	-.1581	.6517	.8077									

TEST 122
 RUN 44
 MACH .745
 R 30.0×10^6

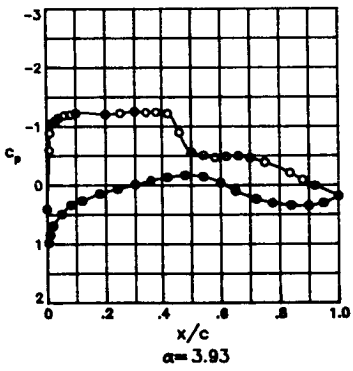
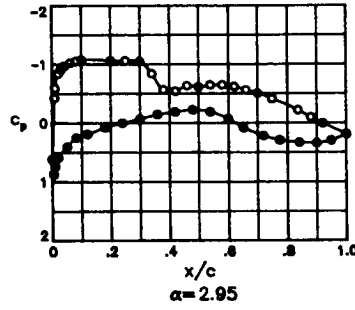
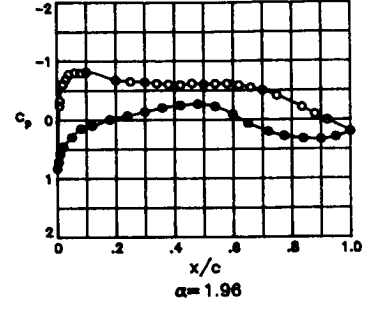
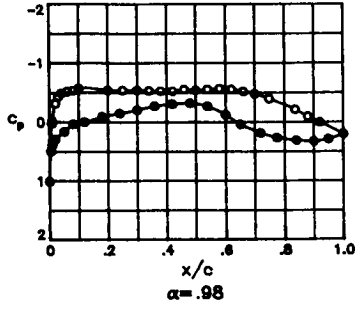
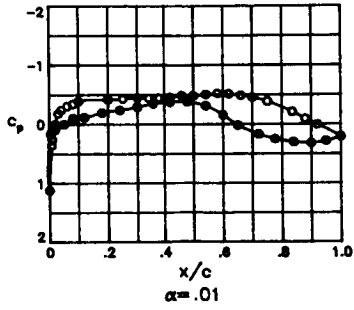


TEST	122	PT	55.4529	PSI	CN	1.0460
RUN	44	TT	112.4744	K	CM	-1.042
POINT	10	RC	29.7270	MILLION	CC	-0.0410
		MACH	.7405			
		ALPHA	5.9187	DEG		

CD1	.04845	CDCDR1	.04816
CD2	.05005	CDCDR2	.04958
CD3	.09724	CDCDR3	.09682
CD4	.03700	CDCDR4	.03669
CD5	.03488	CDCDR5	.03473

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC	X/C	Y/B/2	CP	P _L /PT	MLOC
0.0000	.6275	.7044	.7274	0.0000	.6275	.7044	.7274	.0500	-.3375	-1.2612	.3602	1.3042
.0083	-.8294	.4775	1.0869	.0052	1.0730	.9811	.1656	.3957	-.3375	-1.4649	.3144	1.4024
.0097	-1.2570	.3635	1.2975	.0098	.9560	.9495	.2738	.5009	-.3375	-1.0628	.4120	1.2033
.0203	-1.3630	.3312	1.3651	.0200	.8093	.9092	.3721	.6048	-.3375	-.4493	.5655	.9428
.0300	-1.3713	.3234	1.3623	.0500	.6007	.8542	.4810	.7003	-.3375	-.3881	.5932	.8991
.0400	-1.4493	.3062	1.4211	.0813	.4354	.8084	.5610					
.0608	-1.3953	.3161	1.3986	.1199	.3538	.7858	.5985					
.0800	-1.3818	.3179	1.3945	.1796	.2302	.7585	.6425					
.1000	-1.44928	.3030	1.4284	.2397	.1413	.7311	.6859					
.1997	-1.4151	.3152	1.4005	.2995	.0539	.7080	.7219					
.2500	-1.3874	.3159	1.3991	.3588	-.0366	.6803	.7647					
.2994	-1.4511	.3117	1.4086	.4193	-.0944	.6715	.7782					
.3402	-1.4534	.3148	1.4015	.4793	-1.1403	.6613	.7939					
.3795	-1.4165	.3111	1.4098	.5394	-1.339	.6557	.8025					
.4201	-1.4295	.3096	1.4133	.5994	-.0346	.6834	.7599					
.4598	-1.4415	.3029	1.4287	.6597	.0930	.7160	.7094					
.4996	-1.0703	.4148	1.1982	.7203	.2258	.7574	.6443					
.5397	-.7352	.5067	1.0378	.7743	.2969	.7779	.6114					
.5795	-.6594	.5157	1.0230	.8394	.3220	.7802	.6076					
.6197	-.5330	.5572	.9559	.8996	.3266	.7842	.6010					
.6598	-.4284	.5860	.9105	.9492	.2608	.7676	.6280					
.6997	-.3728	.5972	.8929	1.0000	.1079	.7259	.6939					
.7493	-.3295	.6088	.8748									
.8353	-.1973	.6479	.8145									
.8791	-.1034	.6676	.7840									
.9212	-.0264	.6853	.7569									
1.0000	.1079	.7259	.6939									

TEST 122
RUN 55
MACH .745
R 45.0×10^6



TEST	122	PT	75.7155	PSI	CN	.7197	CD1	.00818	CDCOR1	.00784		
RUN	55	TT	105.1986	K	CM	-.1008	CD2	.00835	CDCOR2	.00803		
POINT	4	PC	44.9440	MILLION	CC	-.0192	CD3	.00808	CDCOR3	.00777		
		MACH	.7425				CD4	.00730	CDCOR4	.00719		
		ALPHA	2.9500	DEG			CD5	.00683	CDCOR5	.00669		
UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC	X/C	Y/B/2	CP	P _L /PT	MLOC
0.0000	.6140	.8579	.4753	0.0000	.6140	.8579	.4753	.0500	-.3375	-.7800	.4893	1.0690
.0003	-.4277	.5800	.9220	.0005	.8712	.9267	.3331	.3957	-.3375	-.5559	.5469	.9745
.0097	-.5944	.5361	.9918	.0098	.7362	.8915	.4103	.5008	-.3375	-.5972	.5331	.9968
.0203	-.8391	.4736	1.0957	.0200	.5896	.8531	.4841	.6048	-.3375	-.6128	.5289	1.0036
.0300	-.9141	.4553	1.1275	.0300	.4073	.8032	.5711	.7003	-.3375	-.4983	.5625	.9495
.0400	-.9714	.4360	1.1617	.0413	.2534	.7625	.6376					
.0608	-1.0245	.4225	1.1863	.1199	.1904	.7477	.6612					
.0800	-1.0515	.4193	1.1921	.1796	.0759	.7158	.7113					
.1000	-1.0750	.4101	1.2092	.2397	-.0009	.6953	.7431					
.1997	-1.0621	.4148	1.2004	.2995	-.0660	.6789	.7686					
.2500	-1.0586	.4144	1.2012	.3588	-.1410	.6582	.8005					
.2994	-1.0473	.4132	1.2033	.4193	-.1869	.6433	.8234					
.3402	-.8376	.4723	1.0980	.4793	-.2227	.6360	.8347					
.3795	-.5635	.5442	.9789	.5394	-.1866	.6446	.8214					
.4201	-.3389	.5523	.9658	.5994	-.0669	.6778	.7703					
.4598	-.0527	.5314	.9994	.6507	.0835	.7203	.7043					
.4996	-.6138	.5320	.9985	.7203	.2198	.7537	.6517					
.5397	-.6451	.5268	1.0069	.7743	.2937	.7750	.6174					
.5795	-.6448	.5262	1.0079	.8394	.3352	.7856	.6001					
.6197	-.6176	.5321	.9983	.8996	.3445	.7874	.5972					
.6598	-.5595	.5493	.9707	.9492	.2995	.7765	.6151					
.6997	-.5066	.5606	.9525	1.0000	.1921	.7470	.6623					
.7493	-.4109	.5848	.9143									
.8353	-.2174	.6406	.8275									
.8791	-.0453	.6722	.7788									
.9212	.0080	.6966	.7412									
1.0000	.1921	.7470	.6623									

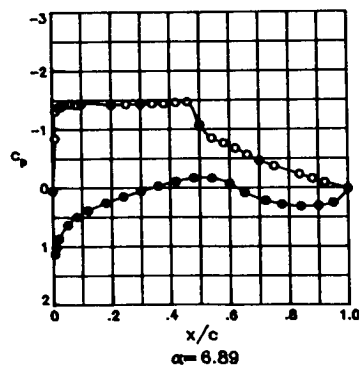
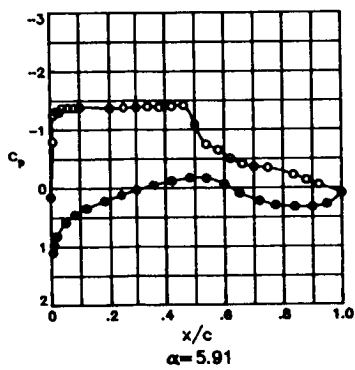
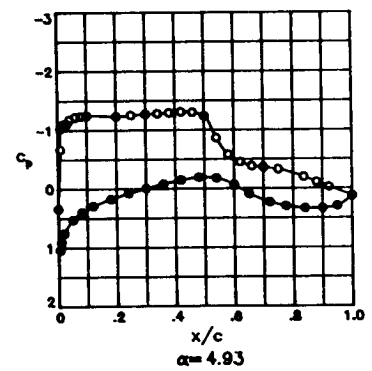
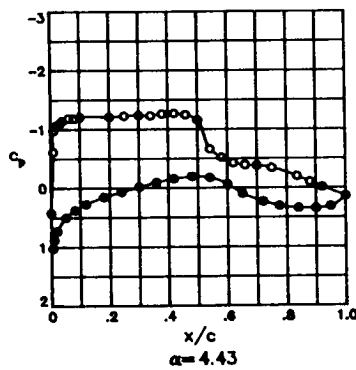
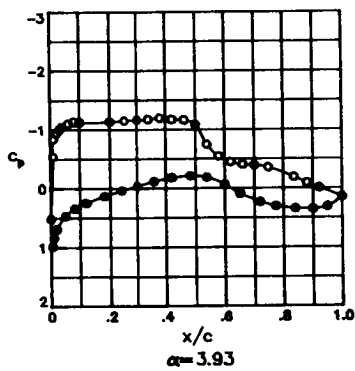
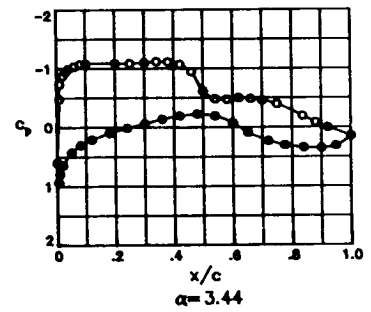
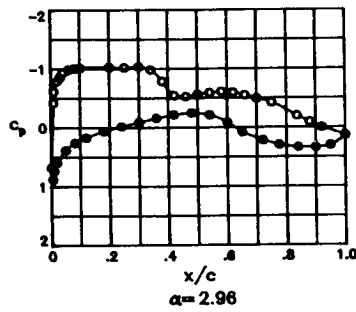
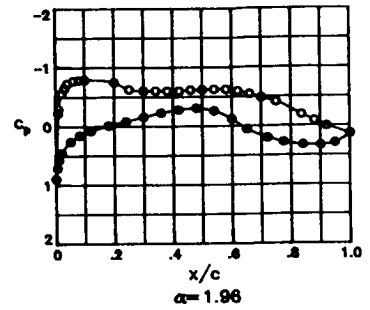
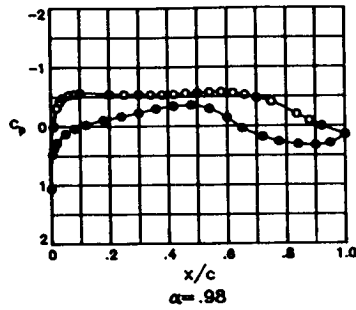
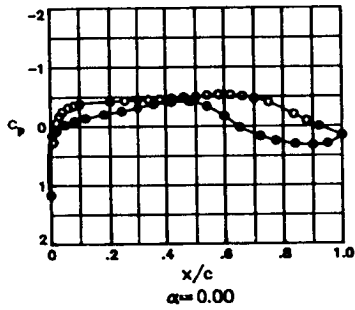
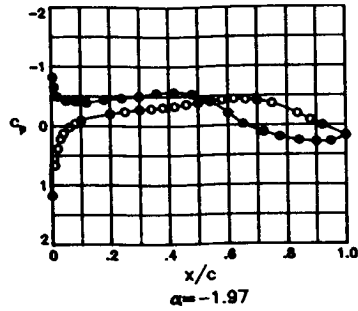
TEST	122	PT	75.7134	PSI	CN	.8679	CD1	.01447	CDCOR1	.01477		
RUN	55	TT	105.2734	K	CM	-.1013	CD2	.01554	CDCOR2	.01579		
POINT	5	PC	44.7900	MILLION	CC	-.0294	CD3	.01458	CDCOR3	.01466		
		MACH	.7404				CD4	.01239	CDCOR4	.01272		
		ALPHA	3.9300	DEG			CD5	.01194	CDCOR5	.01218		
UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC	X/C	Y/B/2	CP	P _L /PT	MLOC
0.0000	.4080	.8027	.5719	0.0000	.4080	.8027	.5719	.0500	-.3375	-.9329	.4513	1.1345
.0043	-.5946	.5350	.9937	.0052	.9807	.9565	.2540	.3957	-.3375	-1.1730	.3879	1.2513
.0097	-.8883	.4614	1.1168	.0098	.8418	.9193	.3503	.5008	-.3375	-.6179	.5318	.9987
.0203	-1.0560	.4153	1.1985	.0200	.6695	.8776	.4378	.6048	-.3375	-.5079	.5648	.9458
.0300	-1.0960	.4603	1.2276	.0300	.4934	.8260	.5322	.7003	-.3375	-.4473	.5770	.9266
.0400	-1.1397	.3910	1.2454	.0413	.3369	.7845	.6020					
.0608	-1.1810	.3804	1.2660	.1199	.2667	.7666	.6311					
.0800	-1.1952	.3783	1.2707	.1796	.1453	.7339	.6832					
.1000	-1.2237	.3690	1.2886	.2397	.0678	.7157	.7114					
.1997	-1.2026	.3707	1.2851	.2995	-.0108	.6898	.7516					
.2500	-1.2214	.3705	1.2856	.3588	-.0786	.6744	.7755					
.2994	-1.2512	.3664	1.2939	.4193	-.1372	.6610	.7961					
.3402	-1.2365	.3686	1.2894	.4793	-.1693	.6515	.8108					
.3795	-1.2394	.3673	1.2921	.5394	-.1462	.6573	.8018					
.4201	-1.2262	.3744	1.2778	.5994	-.0432	.6858	.7579					
.4598	-.8475	.4627	1.1146	.6507	.1012	.7256	.6960					
.4996	-.5806	.5489	.9714	.7203	.2355	.7595	.6425					
.5397	-.5063	.5600	.9536	.7743	.3051	.7760	.6158					
.5795	-.4671	.5706	.9367	.8394	.3426	.7861	.5993					
.6197	-.4481	.5655	.9448	.8996	.3508	.7887	.5951					
.6598	-.4957	.5670	.9424	.9492	.3037	.7781	.6125					
.6997	-.4597	.5754	.9291	1.0000	.1887	.7494	.6585					
.7493	-.3661	.5945	.8991									
.8353	-.2102	.6436	.8230									
.8791	-.0805	.6740	.7760									
.9212	.0056	.6990	.7373									
1.0000	.1987	.7494	.6585									

Appendix E

Pressure Data for $M = 0.76$; $R = 4.4 \times 10^6$, 7.7×10^6 , 14.0×10^6 , 30.0×10^6 , and 45.0×10^6 ; and Free Transition

The pressure measurements made on the Boeing BAC I airfoil are presented in coefficient form in graphs and tables in this appendix. The data are for a Mach number of 0.76; Reynolds numbers of 4.4×10^6 , 7.7×10^6 , 14.0×10^6 , 30.0×10^6 , and 45.0×10^6 ; and free transition. The pressure data from the upper surface of the airfoil are plotted as open symbols, and the data from the lower surface are plotted as solid symbols.

TEST 122
 RUN 20
 MACH .765
 R 4.4×10^6



TEST 122 PT 17.6162 PSI CN .8520
 RUN 20 TT 192.4418 K CM -.1073
 POINT R RC 4.4406 MILLION CC -.0252
 MACH .7635
 ALPHA 3.9300 DEG

CD1 .01721 CDCOR1 .01651
 CD2 .01622 CDCOR2 .01550
 CD3 .01799 CDCOR3 .01723
 CD4 .01602 CDCOR4 .01529
 CD5 .01329 CDCOR5 .01258

UPPER SURFACE
 X/C CP P_L/PT MLOC
 0.0000 .5168 .8225 .5357
 .0083 -.5426 .5277 1.0008
 .0097 -.8444 .4457 1.1395
 .0203 -.9427 .4184 1.1888
 .0300 -1.0041 .4012 1.2209
 .0400 -1.0518 .3900 1.2423
 .0608 -1.1037 .3744 1.2729
 .0800 -1.1369 .3704 1.2809
 .1000 -1.1292 .3655 1.2907
 .1997 -1.1365 .3649 1.2920
 .2500 -1.1513 .3595 1.3029
 .2994 -1.1641 .3543 1.3136
 .3402 -1.1734 .3540 1.3142
 .3795 -1.1939 .3503 1.3218
 .4201 -1.1700 .3503 1.3218
 .4598 -1.1607 .3567 1.3085
 .4996 -1.0444 .3805 1.2608
 .5397 -.7428 .4755 1.0876
 .5795 -.5434 .5277 1.0009
 .6197 -.4444 .5567 .9543
 .6598 -.4014 .5683 .9359
 .6997 -.3874 .5721 .9299
 .7493 -.3509 .5813 .9154
 .8353 -.1848 .6274 .8438
 .8791 -.0876 .6543 .8024
 .9212 -.0014 .6784 .7654
 1.0000 .1464 .7191 .7026

LOWER SURFACE
 X/C CP P_L/PT MLOC
 0.0000 .5168 .8225 .5357
 .0052 .9819 .9522 .2652
 .0098 .8346 .9114 .3663
 .0200 .6837 .8698 .4513
 .0500 .4657 .8098 .5572
 .0813 .3381 .7739 .6161
 .1199 .2440 .7503 .6539
 .1796 .1302 .7154 .7083
 .2397 .0379 .6896 .7482
 .2995 -.0381 .6694 .7792
 .3588 -.1189 .6462 .8148
 .4193 -.1815 .6278 .8431
 .4793 -.2141 .6202 .8549
 .5394 -.1842 .6297 .8402
 .5994 -.0655 .6589 .7955
 .6507 .0959 .7059 .7231
 .7203 .2345 .7456 .6612
 .7743 .3078 .7661 .6287
 .8394 .3484 .7757 .6132
 .8996 .3546 .7783 .6091
 .9492 .3171 .7676 .6262
 1.0000 .1464 .7191 .7026

SPANWISE
 X/C Y/B/2 CP P_L/PT MLOC
 .0500 -.3375 -.8444 .4474 1.1364
 .3957 -.3375 -1.1242 .3693 1.2831
 .5008 -.3375 -1.0366 .3962 1.2305
 .6048 -.3375 -.4641 .5523 .9612
 .7003 -.3375 -.3841 .5698 .9334

TEST 122 PT 17.6164 PSI CN .9001
 RUN 20 TT 192.7224 K CM -.1080
 POINT 9 RC 4.4106 MILLION CC -.0289
 MACH .7572
 ALPHA 4.4293 DEG

CD1 .02326 CDCOR1 .02249
 CD2 .02325 CDCOR2 .02247
 CD3 .02308 CDCOR3 .02231
 CD4 .01901 CDCOR4 .01850
 CD5 .01637 CDCOR5 .01608

UPPER SURFACE
 X/C CP P_L/PT MLOC
 0.0000 .4218 .7975 .5776
 .0083 -.6169 .5101 1.0297
 .0097 -.9817 .4134 1.1981
 .0203 -1.0668 .3925 1.2376
 .0300 -1.0950 .3809 1.2601
 .0400 -1.1461 .3680 1.2856
 .0608 -1.1863 .3576 1.3067
 .0800 -1.1826 .3527 1.3169
 .1000 -1.2078 .3510 1.3203
 .1997 -1.2090 .3501 1.3222
 .2500 -1.2270 .3466 1.3297
 .2994 -1.2391 .3414 1.3406
 .3402 -1.2314 .3396 1.3444
 .3795 -1.2610 .3377 1.3486
 .4201 -1.2689 .3358 1.3525
 .4598 -1.2444 .3400 1.3436
 .4996 -1.1521 .3665 1.2887
 .5397 -.6607 .5038 1.0401
 .5795 -.5155 .5396 .9816
 .6197 -.4249 .5678 .9366
 .6598 -.3879 .5742 .9265
 .6997 -.3792 .5799 .9176
 .7493 -.3387 .5876 .9054
 .8353 -.1945 .6301 .8396
 .8791 -.0989 .6569 .7985
 .9212 -.0088 .6824 .7593
 1.0000 .1394 .7203 .7008

LOWER SURFACE
 X/C CP P_L/PT MLOC
 0.0000 .4218 .7975 .5776
 .0052 1.0215 .9642 .2288
 .0098 .8798 .9237 .3337
 .0200 .7268 .8827 .4257
 .0500 .5010 .8210 .5382
 .0813 .3732 .7861 .5964
 .1199 .2706 .7553 .6459
 .1796 .1507 .7246 .6940
 .2397 .0661 .7011 .7304
 .2995 -.0194 .6775 .7668
 .3588 -.0946 .6576 .7973
 .4193 -.1565 .6394 .8254
 .4793 -.1945 .6267 .8449
 .5394 -.1751 .6358 .8309
 .5994 -.0635 .6666 .7936
 .6507 .0995 .7101 .7166
 .7203 .2346 .7478 .6578
 .7743 .3082 .7692 .6238
 .8394 .3454 .7772 .6108
 .8996 .3490 .7800 .6062
 .9492 .3149 .7684 .6250
 1.0000 .1394 .7203 .7008

SPANWISE
 X/C Y/B/2 CP P_L/PT MLOC
 .0500 -.3375 -.9210 .4320 1.1640
 .3957 -.3375 -1.2059 .3558 1.3105
 .5008 -.3375 -1.0260 .4019 1.2197
 .6048 -.3375 -.4493 .5600 .9490
 .7003 -.3375 -.3845 .5773 .9216

TEST 122 PT 17.6161 PSI CN .9372
 RUN 20 TT 192.8849 K CM -.1085
 POINT 11 RC 4.4207 MILLION CC -.0302
 MACH .7615
 ALPHA 4.9326 DEG

CD1 .03147 CDCOR1 .03029
 CD2 .03221 CDCOR2 .03094
 CD3 .03099 CDCOR3 .02975
 CD4 .02451 CDCOR4 .02350
 CD5 .02060 CDCOR5 .01985

UPPER SURFACE
 X/C CP P_L/PT MLOC
 0.0000 .3357 .7752 .6140
 .0083 -.4738 .4973 1.0508
 .0097 -1.0287 .3941 1.2344
 .0203 -1.1033 .3741 1.2734
 .0300 -1.1083 .3690 1.2836
 .0400 -1.1793 .3529 1.3165
 .0608 -1.2287 .3440 1.3350
 .0800 -1.2393 .3381 1.3475
 .1000 -1.2502 .3363 1.3516
 .1997 -1.2342 .3369 1.3502
 .2500 -1.2553 .3326 1.3594
 .2994 -1.2728 .3283 1.3688
 .3402 -1.2916 .3283 1.3689
 .3795 -1.2933 .3243 1.3776
 .4201 -1.3070 .3216 1.3837
 .4598 -1.2989 .3178 1.3920
 .4996 -1.2364 .3407 1.3421
 .5397 -.8622 .4402 1.1492
 .5795 -.5803 .5210 1.0118
 .6197 -.4538 .5545 .9577
 .6598 -.3808 .5764 .9231
 .6997 -.3622 .5839 .9112
 .7493 -.3200 .5855 .9041
 .8353 -.1965 .6255 .8467
 .8791 -.1067 .6521 .8057
 .9212 -.0239 .6738 .7725
 1.0000 .1263 .7141 .7103

LOWER SURFACE
 X/C CP P_L/PT MLOC
 0.0000 .3357 .7752 .6140
 .0052 1.0386 .9679 .2164
 .0098 .9036 .9306 .3221
 .0200 .7496 .8869 .4175
 .0500 .5239 .8252 .5309
 .0813 .3937 .7910 .5884
 .1199 .2909 .7614 .6361
 .1796 .1739 .7296 .6863
 .2397 .0773 .7019 .7292
 .2995 -.0073 .6775 .7668
 .3588 -.0856 .6566 .7988
 .4193 -.1478 .6397 .8248
 .4793 -.1914 .6291 .8412
 .5394 -.1774 .6325 .8360
 .5994 -.0659 .6639 .7877
 .6507 .0949 .7053 .7246
 .7203 .2306 .7455 .6613
 .7743 .3017 .7633 .6331
 .8394 .3387 .7749 .6145
 .8996 .3383 .7740 .6160
 .9492 .2984 .7640 .6321
 1.0000 .1263 .7141 .7103

SPANWISE
 X/C Y/B/2 CP P_L/PT MLOC
 .0500 -.3375 -.9131 .4258 1.1752
 .3957 -.3375 -1.2287 .3390 1.3458
 .5008 -.3375 -1.1920 .3519 1.3185
 .6048 -.3375 -.4857 .5485 .9672
 .7003 -.3375 -.3677 .5793 .9185

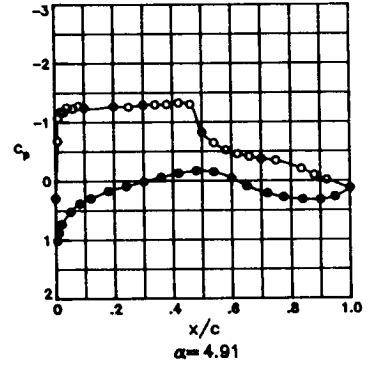
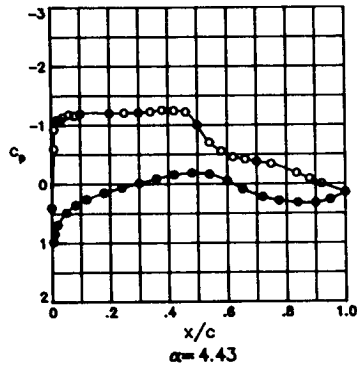
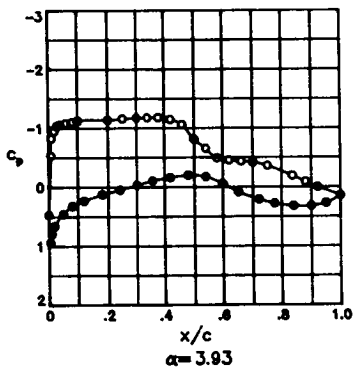
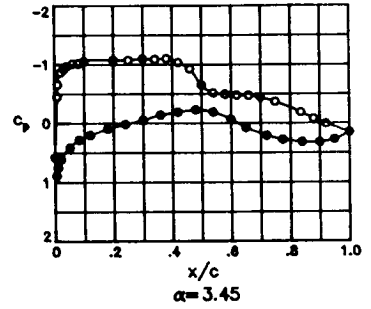
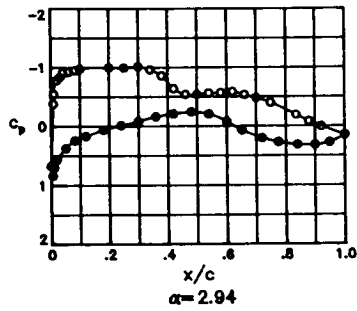
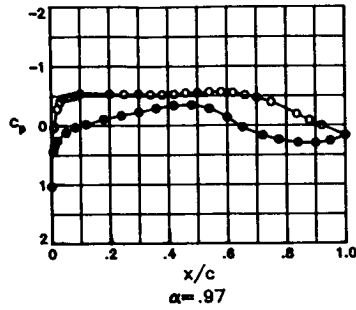
TEST 122	PT 17.6158	PSI	CN 1.0176	CD1 .04908	CDCOR1 .04797
RUN 20	TT 192.8211	K	CM -.1070	CD2 .05088	CDCOR2 .04978
POINT 12	RC 4.3998	MILLION	CC -.0359	CD3 .04671	CDCOR3 .04559
	MACH .7548			CD4 .03234	CDCOR4 .03136
	ALPHA 5.9090	DEG		CD5 .02688	CDCOR5 .02618

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P ₂ L/PT	MLOC	X/C	CP	P ₂ L/PT	MLOC	X/C	Y/B/2	CP	P ₂ L/PT	MLOC
0.0000	.1499	.7274	.6896	0.0000	.1499	.7274	.6896	.0500	-.3375	-1.0239	.4048	1.2142
.0083	-.7992	.4687	1.0993	.0052	1.1004	.9868	.1377	.3957	-.3375	-1.3728	.3096	1.4106
.0097	-1.2475	.3490	1.3246	.0098	.9733	.9519	.2661	.5008	-.3375	-.9857	.4136	1.1978
.0203	-1.3144	.3286	1.3682	.0200	.8165	.9078	.3743	.6048	-.3375	-.5071	.5463	.9709
.0300	-1.3023	.3250	1.3761	.0300	.5865	.8455	.4954	.7003	-.3375	-.3730	.5821	.9141
.0400	-1.3715	.3094	1.4112	.0813	.4515	.8092	.5582					
.0508	-1.3771	.3097	1.4104	.1199	.3419	.7785	.6087					
.0800	-1.3733	.3088	1.4124	.1796	.2181	.7454	.6616					
.1000	-1.3859	.3072	1.4162	.2397	.1180	.7177	.7047					
.1997	-1.3704	.3115	1.4064	.2995	-.0336	.6950	.7399					
.2500	-1.3768	.3090	1.4120	.3588	-.0512	.6714	.7761					
.2994	-1.3834	.3061	1.4180	.4193	-.1212	.6517	.8064					
.3402	-1.3959	.3057	1.4196	.4793	-.1724	.6394	.8254					
.3795	-1.3852	.3030	1.4259	.5394	-.1674	.6376	.8282					
.4201	-1.4023	.2994	1.4343	.5994	-.0661	.6659	.7845					
.4598	-1.4121	.2985	1.4363	.6597	.0915	.7101	.7166					
.4996	-1.0942	.3870	1.2482	.7203	.2247	.7459	.6607					
.5397	-.7467	.4816	1.0772	.7743	.2979	.7670	.6272					
.5795	-.6501	.5060	1.0364	.8394	.3288	.7744	.6153					
.6197	-.5022	.5475	.9688	.8996	.3272	.7746	.6151					
.6598	-.4089	.5734	.9278	.9492	.2828	.7626	.6342					
.6997	-.3595	.5877	.9054	1.0000	.0921	.7099	.7169					
.7493	-.3392	.5944	.8950									
.8353	-.2190	.6262	.8456									
.8791	-.1327	.6491	.8104									
.9212	-.0533	.6692	.7796									
1.0000	.0921	.7099	.7169									

TEST 122	PT 17.6136	PSI	CN 1.0850	CD1 .07374	CDCOR1 .07278
RUN 20	TT 192.7749	K	CM -.1158	CD2 .07469	CDCOR2 .07357
POINT 13	RC 4.4175	MILLION	CC -.0350	CD3 .07150	CDCOR3 .07030
	MACH .7598			CD4 .04623	CDCOR4 .04536
	ALPHA 6.8926	DEG		CD5 .03427	CDCOR5 .03371

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P ₂ L/PT	MLOC	X/C	CP	P ₂ L/PT	MLOC	X/C	Y/B/2	CP	P ₂ L/PT	MLOC
0.0000	.0523	.6965	.7375	0.0000	.0523	.6965	.7375	.0500	-.3375	-1.0609	.3901	1.2421
.0083	-.8463	.4487	1.1342	.0052	1.1313	.9943	.0900	.3957	-.3375	-1.4588	.2601	1.4809
.0097	-1.3102	.3231	1.3801	.0098	1.0174	.9627	.2335	.5008	-.3375	-1.0633	.3894	1.2435
.0203	-1.3781	.3021	1.4280	.0200	.8651	.9203	.3483	.6048	-.3375	-.8634	.4971	1.0511
.0300	-1.3921	.2963	1.4415	.0300	.6330	.8567	.4751	.7003	-.3375	-.4285	.5612	.9470
.0400	-1.4265	.2888	1.4595	.0813	.4930	.8173	.5445					
.0508	-1.4325	.2846	1.4698	.1199	.3830	.7862	.5962					
.0800	-1.4204	.2862	1.4658	.1796	.2535	.7512	.6523					
.1000	-1.4405	.2828	1.4741	.2397	.1493	.7228	.6969					
.1997	-1.4208	.2905	1.4555	.2995	.0588	.6984	.7346					
.2500	-1.4289	.2867	1.4646	.3588	-.0289	.6735	.7730					
.2994	-1.4308	.2841	1.4710	.4193	-.1086	.6503	.8085					
.3402	-1.4392	.2841	1.4708	.4793	-.1679	.6352	.8318					
.3795	-1.4372	.2811	1.4783	.5394	-.1632	.6345	.8328					
.4201	-1.4563	.2805	1.4798	.5994	-.0745	.6616	.7913					
.4598	-1.4688	.2785	1.4848	.6597	.0861	.7065	.7221					
.4996	-1.0755	.3876	1.2470	.7203	.2185	.7433	.6647					
.5397	-.8467	.4469	1.1373	.7743	.2848	.7599	.6387					
.5795	-.7647	.4743	1.0897	.8394	.3183	.7714	.6201					
.6197	-.6763	.4937	1.0568	.8996	.3067	.7657	.6293					
.6598	-.5635	.5264	1.0029	.9492	.2559	.7525	.6503					
.6997	-.4644	.5527	.9605	1.0000	.0141	.6843	.7564					
.7493	-.3704	.5791	.9188									
.8353	-.2289	.6184	.8577									
.8791	-.1531	.6404	.8238									
.9212	-.0910	.6571	.7982									
1.0000	.0141	.6843	.7564									

TEST 122
RUN 28
MACH .765
R 7.7×10^6



ORIGINAL PAGE IS
OF POOR QUALITY

TEST 122 PT 17.6922 PSI CN .8243
RUN 28 TT 130.4284 K CM -.0988
POINT 7 RC 7.7944 MILLION CC -.0255
MACH .7628
ALPHA 3.9300 DEG

CD1 .01856 CDCOR1 .01769
CD2 .01790 CDCOR2 .01696
CD3 .01843 CDCOR3 .01764
CD4 .01695 CDCOR4 .01622
CD5 .01420 CDCOR5 .01372

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC
0.0000	.4956	.8174	.5447	0.0000	.4956	.8174	.5447
.0083	-.5180	.5364	.9873	.0052	.9433	.9413	.2952
-.0097	-.7792	.4632	1.1093	.0098	.8036	.9035	.3836
.0203	-.9751	.4422	1.2007	.0200	.6811	.8644	.4611
.0300	-1.0178	.4015	1.2209	.0500	.4603	.8094	.5583
.0400	-1.0595	.3908	1.2413	.0813	.3230	.7703	.6224
.0608	-1.0802	.3822	1.2586	.1199	.2336	.7469	.6596
.0800	-1.0946	.3810	1.2603	.1796	.1216	.7108	.7160
.1000	-1.0866	.3731	1.2759	.2397	.0463	.6922	.7447
.1997	-1.1470	.3663	1.2895	.2995	-.0242	.6757	.7700
.2500	-1.1234	.3633	1.2957	.3588	-.1102	.6462	.8153
.2994	-1.1499	.3579	1.3065	.4193	-.1666	.6318	.8375
.3402	-1.1596	.3576	1.3073	.4793	-.1979	.6246	.8487
.3795	-1.1794	.3549	1.3128	.5394	-.1766	.6322	.8370
.4201	-1.1812	.3586	1.3052	.5994	-.0606	.6666	.7839
.4598	-1.1330	.3692	1.2836	.6507	.0873	.7059	.7234
.4996	-.9444	.4214	1.1838	.7203	.2155	.7414	.6683
.5397	-.5662	.5298	.9980	.7743	.2829	.7623	.6352
.5795	-.3371	.5274	1.0018	.8394	.3232	.7675	.6268
.6197	-.4596	.5923	.9617	.8996	.3243	.7697	.6232
.6598	-.4263	.5643	.9426	.9492	.2732	.7573	.6431
.6997	-.3916	.5688	.9356	1.0000	.1482	.7206	.7007
.7493	-.3497	.5845	.9109				
.8353	-.1735	.6280	.8434				
.8791	-.0802	.6566	.7995				
.9212	.0067	.6821	.7602				
1.0000	.1482	.7206	.7007				

SPANWISE				
X/C	Y/B/2	CP	P/L/PT	MLOC
.0500	-.3375	-.9490	.4158	1.1941
.3957	-.3375	-1.1139	.3731	1.2759
.5008	-.3375	-1.0749	.3812	1.2599
.6048	-.3375	-.4744	.5471	.9701
.7003	-.3375	-.4068	.5688	.9356

TEST 122 PT 17.6708 PSI CN .8771
RUN 28 TT 130.4075 K CM -.0977
POINT 8 RC 7.7578 MILLION CC -.0799
MACH .7579
ALPHA 4.4100 DEG

CD1 .02412 CDCOR1 .02345
CD2 .02510 CDCOR2 .02431
CD3 .02361 CDCOR3 .02289
CD4 .02002 CDCOR4 .01952
CD5 .01762 CDCOR5 .01726

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC
0.0000	.3767	.7872	.5950	0.0000	.3767	.7872	.5950
.0083	-.6123	.5156	1.0211	.0052	.9831	.9541	.2600
-.0097	-.9775	.4177	1.1906	.0098	.8436	.9150	.3584
.0203	-1.0640	.3902	1.2424	.0200	.6987	.8757	.4396
.0300	-1.1063	.3804	1.2614	.0300	.4897	.8188	.5423
.0400	-1.1685	.3646	1.2930	.0813	.3537	.7814	.6044
.0608	-1.1703	.3638	1.2946	.1199	.2585	.7559	.6453
.0800	-1.1843	.3612	1.2998	.1796	.1484	.7246	.6945
.1000	-1.1945	.3559	1.3107	.2397	.0685	.7008	.7314
.1997	-1.2060	.3521	1.3186	.2995	-.0125	.6800	.7634
.2500	-1.2044	.3500	1.3229	.3588	-.0931	.6564	.7996
.2994	-1.2331	.3450	1.3335	.4193	-.1457	.6436	.8193
.3402	-1.2450	.3450	1.3333	.4793	-.1795	.6363	.8305
.3795	-1.2450	.3403	1.3433	.5394	-.1615	.6385	.8272
.4201	-1.2412	.3378	1.3497	.5994	-.0516	.6665	.7842
.4598	-1.2258	.3416	1.3410	.6507	.0826	.7035	.7272
.4996	-.9798	.4131	1.1991	.7203	.2131	.7414	.6681
.5397	-.6251	.5106	1.0293	.7743	.2789	.7595	.6396
.5795	-.5191	.5463	.9810	.8394	.3217	.7716	.6203
.6197	-.4374	.5551	.9572	.8996	.3136	.7648	.6311
.6598	-.4138	.5709	.9323	.9492	.2654	.7571	.6434
.6997	-.3761	.5774	.9220	1.0000	.1408	.7206	.7007
.7493	-.3419	.5892	.9035				
.8353	-.1839	.6336	.8347				
.8791	-.0862	.6685	.7919				
.9212	.0016	.6937	.7577				
1.0000	.1408	.7206	.7007				

SPANWISE				
X/C	Y/B/2	CP	P/L/PT	MLOC
.0500	-.3375	-1.0457	.3975	1.2284
.3957	-.3375	-1.1851	.3587	1.3051
.5008	-.3375	-.9736	.4150	1.1955
.6048	-.3375	-.4605	.5561	.9557
.7003	-.3375	-.3856	.5756	.9247

TEST 122 PT 17.5762 PSI CN .9142
RUN 28 TT 130.3852 K CM -.1025
POINT 9 RC 7.7299 MILLION CC -.0306
MACH .7627
ALPHA 4.9100 DEG

CD1 .03232 CDCOR1 .03185
CD2 .03452 CDCOR2 .03392
CD3 .03143 CDCOR3 .03083
CD4 .02514 CDCOR4 .02479
CD5 .02086 CDCOR5 .02072

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC
0.0000	.3313	.7730	.6179	0.0000	.3313	.7730	.6179
.0083	-.6496	.5022	1.0432	.0052	1.0130	.9620	.2360
-.0097	-1.0298	.4015	1.2207	.0098	.8827	.9265	.3321
.0203	-1.1355	.3685	1.2851	.0200	.7267	.8822	.4270
.0300	-1.1282	.3700	1.2820	.0500	.5119	.8206	.5392
.0400	-1.1607	.3545	1.3136	.0813	.3786	.7866	.5960
.0608	-1.2039	.3503	1.3223	.1199	.2777	.7559	.6453
.0800	-1.1955	.3460	1.3312	.1796	.1651	.7255	.6931
.1000	-1.2097	.3440	1.3355	.2397	.0824	.7005	.7319
.1997	-1.2404	.3409	1.3420	.2995	.0060	.6842	.7569
.2500	-1.2446	.3360	1.3524	.3588	-.0754	.6597	.7946
.2994	-1.2414	.3303	1.3649	.4193	-.1510	.6349	.8328
.3402	-1.2720	.3321	1.3610	.4793	-.1849	.6316	.8379
.3795	-1.2759	.3277	1.3706	.5394	-.1622	.6353	.8320
.4201	-1.2994	.3254	1.3766	.5994	-.0600	.6664	.7843
.4598	-1.3061	.3235	1.3797	.6507	.0838	.7050	.7234
.4996	-1.0789	.3810	1.2604	.7203	.2043	.7366	.6756
.5397	-.7142	.4870	1.0687	.7743	.2784	.7598	.6391
.5795	-.5999	.5147	1.0225	.8394	.3130	.7673	.6271
.6197	-.4961	.5455	.9726	.8996	.3172	.7696	.6233
.6598	-.4159	.5672	.9380	.9492	.2670	.7542	.6479
.6997	-.3780	.5745	.9266	1.0000	.1133	.7098	.7174
.7493	-.3389	.5869	.9071				
.8353	-.1904	.6252	.8477				
.8791	-.0978	.6529	.8050				
.9212	-.0205	.6749	.7713				
1.0000	.1133	.7098	.7174				

SPANWISE				
X/C	Y/B/2	CP	P/L/PT	MLOC
.0500	-.3375	-1.0874	.3807	1.2610
.3957	-.3375	-1.2199	.3439	1.3357
.5008	-.3375	-1.0724	.3838	1.2548
.6048	-.3375	-.4857	.5469	.9704
.7003	-.3375	-.3855	.5754	.9252

TEST 122 PT 17.6093 PSI CM .8251 CD1 .01782 CDCOR1 .01716
RUN 28 TT 130.2151 K CM -.0986 CD2 .01765 CDCOR2 .01689
POINT 13 RC 7.7400 MILLION CC -.0259 CD3 .01777 CDCOR3 .01677
MACH .7595 CD4 .01563 CDCOR4 .01515
ALPHA 3.9300 DEG CD5 .01338 CDCOR5 .01311
UPPER SURFACE LOWER SURFACE SPANWISE
X/C CP P/L/PT MLOC X/C CP P/L/PT MLOC X/C Y/B/2 CP P/L/PT MLOC

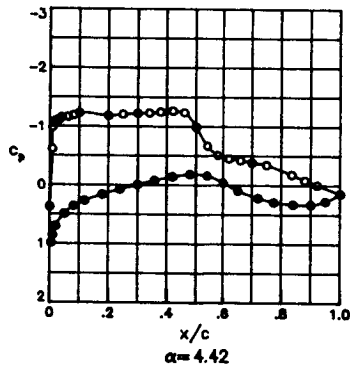
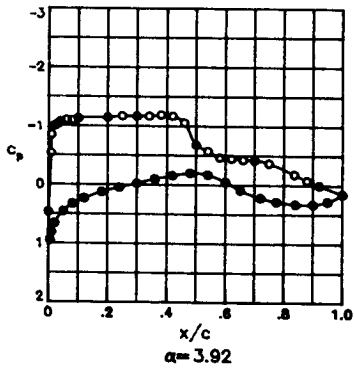
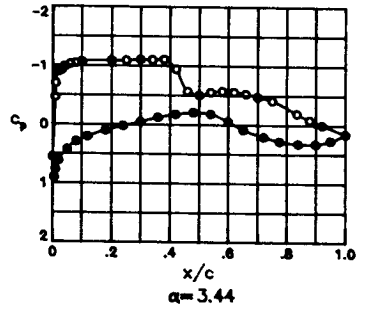
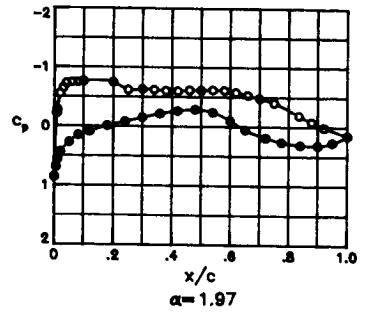
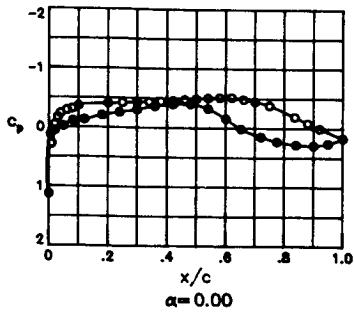
TEST 122 PT 17.5611 PSI CM .7570 CD1 .01328 CDCOR1 .01267
RUN 28 TT 129.9384 K CM -.0951 CD2 .01309 CDCOR2 .01225
POINT 14 RC 7.7641 MILLION CC -.0219 CD3 .01317 CDCOR3 .01258
MACH .7629 CD4 .01208 CDCOR4 .01162
ALPHA 3.4500 DEG CD5 .01098 CDCOR5 .01029
UPPER SURFACE LOWER SURFACE SPANWISE
X/C CP P/L/PT MLOC X/C CP P/L/PT MLOC X/C Y/B/2 CP P/L/PT MLOC

TEST 122 PT 17.5372 PSI CM .6796 CD1 .00998 CDCOR1 .00955
RUN 28 TT 130.2816 K CM -.0947 CD2 .01011 CDCOR2 .00964
POINT 15 RC 7.7264 MILLION CC -.0168 CD3 .01002 CDCOR3 .00960
MACH .7625 CD4 .00867 CDCOR4 .00837
ALPHA 2.9400 DEG CD5 .00743 CDCOR5 .00726
UPPER SURFACE LOWER SURFACE SPANWISE
X/C CP P/L/PT MLOC X/C CP P/L/PT MLOC X/C Y/B/2 CP P/L/PT MLOC

TEST	122	PT	17.0146	PSI		CN	-3946	CD1	.00809	CDCOR1	.00799
RUN	28	TT	131.7951	K		CM	-.0975	CD2	.00806	CDCOR2	.00795
PDINT	16	RC	7.3137	MILLION		CC	.0011	CD3	.00803	CDCOR3	.00790
		MACH	.7516					CD4	.00795	CDCOR4	.00786
		ALPHA	.9700	DEG				CD5	.00707	CDCOR5	.00702

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/2	CP	P/L/PT	MLOC
0.0000	1.0290	.9660	.2229	0.0000	1.0290	.9660	.2229	.0500	-.3375	-.4534	.5692	.9348
.0083	.0392	.6935	.7427	.0052	.4429	.8053	.5650	.3957	-.3375	-.5034	.5581	.9524
.0097	.0162	.6881	.7509	.0098	.3470	.7779	.6101	.5008	-.3375	-.5220	.5543	.9586
.0203	-.2785	.6055	.8782	.0200	.2584	.7546	.6475	.6048	-.3375	-.5261	.5519	.9622
.0300	-.3969	.5745	.9266	.0500	.1205	.7150	.7093	.7003	-.3375	-.4678	.5531	.9604
.0400	-.4594	.5550	.9574	.0813	.0312	.6913	.7461					
.0608	-.4807	.5504	.9648	.1199	-.0194	.6764	.7689					
.0800	-.5000	.5433	.9754	.1794	-.1047	.6538	.8036					
.1000	-.5371	.5348	.9897	.2397	-.1665	.6368	.8298					
.1997	-.5300	.5370	.9862	.2995	-.2273	.6203	.8552					
.2500	-.5262	.5373	.9858	.3588	-.2891	.6027	.8826					
.2994	-.5354	.5363	.9874	.4193	-.3359	.5911	.9005					
.3402	-.5237	.5388	.9834	.4793	-.3445	.5880	.9053					
.3795	-.5225	.5387	.9831	.5394	-.2840	.6046	.8796					
.4201	-.5283	.5368	.9865	.5994	-.1330	.6457	.8161					
.4598	-.5508	.5292	.9988	.6507	.0307	.6899	.7482					
.4996	-.5500	.5316	.9950	.7203	.1683	.7293	.6872					
.5397	-.5650	.5274	1.0018	.7743	.2425	.7496	.6553					
.5795	-.5643	.5265	1.0033	.8394	.2932	.7629	.6341					
.6197	-.5524	.5308	.9963	.8996	.3019	.7659	.6293					
.6598	-.5127	.5425	.9775	.9492	.2634	.7557	.6455					
.6997	-.4492	.5552	.9571	1.0000	.1714	.7378	.6738					
.7493	-.3931	.5750	.9257									
.8353	-.1923	.6293	.8414									
.8791	-.0844	.6648	.7867									
.9212	.0031	.6914	.7458									
1.0000	.1714	.7378	.6738									

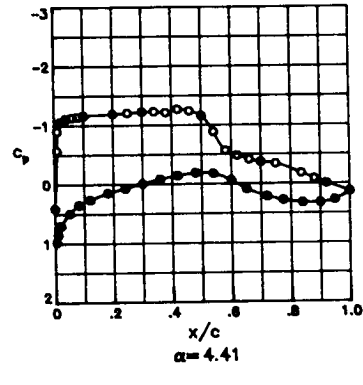
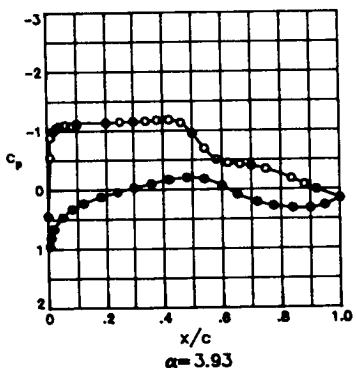
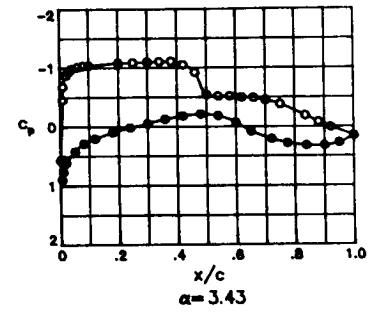
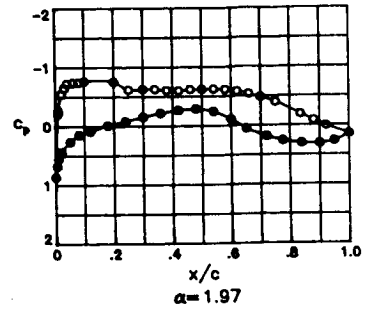
TEST 122
RUN 30
MACH .765
R 7.7×10^6



TEST	122	PT	17.6365	PSI	CN	.8227	CD1	.01764	CDCDR1	.01682		
RUN	30	TT	130.4259	K	CM	-.0968	CD2	.01773	CDCDR2	.01690		
POINT	4	RC	7.7247	MILLION	CC	-.0263	CD3	.01729	CDCDR3	.01645		
		MACH	.7578				CD4	.01517	CDCDR4	.01459		
		ALPHA	3.9200	DEG			CD5	.01353	CDCDR5	.01319		
UPPER SURFACE												
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	SPANWISE				
0.0000	.4689	.8114	.5548	0.0000	.4689	.8114	.5548	X/C	Y/B/Z	CP	P/L/PT	MLOC
.0083	-.5330	.5351	.9893	.0052	.9486	.9439	.2883	.0500	-.3375	-.9999	.4103	1.2042
.0097	-.8477	.4497	1.1329	.0098	.8097	.9064	.3774	.3957	-.3375	-1.1194	.3816	1.2592
.0203	-.9490	.4109	1.2033	.0200	.6604	.8643	.4613	.5008	-.3375	-.8193	.4365	1.2210
.0300	-1.0305	.3984	1.2268	.0500	.4593	.8095	.5580	.6048	-.3375	-.4704	.5503	.9650
.0400	-1.0729	.3812	1.2600	.0813	.3273	.7734	.6173	.7003	-.3375	-.3997	.5692	.9349
.0608	-1.1004	.3778	1.2665	.1199	.2352	.7462	.6606					
.0800	-1.0979	.3778	1.2665	.1796	.1272	.7171	.7061					
.1000	-1.1290	.3708	1.2806	.2397	.0481	.6957	.7392					
.1997	-1.1444	.3672	1.2878	.2995	-.0234	.6760	.7696					
.2500	-1.1700	.3648	1.2925	.3588	-.1012	.6573	.7982					
.2994	-1.1693	.3593	1.3038	.4193	-.1587	.6381	.8278					
.3402	-1.1714	.3583	1.3058	.4793	-.2030	.6256	.8470					
.3795	-1.1916	.3563	1.3099	.5394	-.1734	.6360	.8311					
.4201	-1.1695	.3588	1.3048	.5994	-.0541	.6667	.7838					
.4598	-1.0589	.3974	1.2287	.6507	.0880	.7102	.7168					
.4996	-.6839	.4494	1.0479	.7203	.2164	.7451	.6624					
.5397	-.5757	.5226	1.0696	.7743	.2831	.7597	.6392					
.5795	-.4693	.5564	.9552	.8394	.3236	.7735	.6172					
.6197	-.4519	.5595	.9503	.8996	.3259	.7731	.6178					
.6598	-.4325	.5628	.9450	.9492	.2791	.7590	.6403					
.6997	-.4244	.5695	.9344	1.0000	.1515	.7248	.6942					
.7493	-.3711	.5861	.9082									
.8353	-.1447	.6323	.8367									
.8791	-.0802	.6600	.7942									
.9212	.0036	.6867	.7531									
1.0000	.1915	.7248	.6942									

TEST	122	PT	17.6765	PSI	CN	.8879	CD1	.02392	CDCDR1	.02326		
RUN	30	TT	130.2169	K	CM	-.1007	CD2	.02502	CDCDR2	.02424		
POINT	5	RC	7.7510	MILLION	CC	-.0300	CD3	.02300	CDCDR3	.02224		
		MACH	.7584				CD4	.01963	CDCDR4	.01910		
		ALPHA	4.4196	DEG			CD5	.01752	CDCDR5	.01719		
UPPER SURFACE												
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	SPANWISE				
0.0000	.3699	.7853	.5980	0.0000	.3699	.7853	.5980	X/C	Y/B/Z	CP	P/L/PT	MLOC
.0083	-.6112	.5160	1.0204	.0052	.9848	.9542	.2597	.0500	-.3375	-1.0686	.3699	1.2430
.0097	-.9830	.4142	1.1971	.0098	.8521	.9185	.3507	.3957	-.3375	-1.2038	.3594	1.3035
.0203	-1.0416	.3875	1.2476	.0200	.6996	.8758	.4396	.5008	-.3375	-1.0423	.3939	1.2353
.0300	-1.1096	.3788	1.2647	.0500	.4924	.8178	.5440	.6048	-.3375	-.4594	.5587	.9515
.0400	-1.1513	.3645	1.2933	.0813	.3568	.7802	.6063	.7003	-.3375	-.3906	.5766	.9234
.0608	-1.1639	.3606	1.3011	.1199	.2671	.7577	.6423					
.0800	-1.1901	.3586	1.3052	.1796	.1614	.7316	.6834					
.1000	-1.2290	.3540	1.3147	.2397	.0731	.7016	.7302					
.1997	-1.1833	.3505	1.3218	.2995	-.0082	.6770	.7681					
.2500	-1.2079	.3481	1.3269	.3588	-.0830	.6587	.7961					
.2994	-1.2260	.3450	1.3335	.4193	-.1446	.6428	.8206					
.3402	-1.2298	.3441	1.3352	.4793	-.1855	.6316	.8377					
.3795	-1.2496	.3408	1.3424	.5394	-.1616	.6394	.8257					
.4201	-1.2651	.3387	1.3468	.5994	-.0448	.6727	.7746					
.4598	-1.2376	.3496	1.3238	.6507	.0919	.7118	.7144					
.4996	-.9440	.4128	1.1997	.7203	.2164	.7441	.6639					
.5397	-.6702	.5013	1.0447	.7743	.2894	.7635	.6332					
.5795	-.5099	.5447	.9739	.8394	.3255	.7731	.6178					
.6197	-.4530	.5647	.9420	.8996	.3292	.7772	.6111					
.6598	-.4237	.5723	.9300	.9492	.2717	.7613	.6366					
.6997	-.3866	.5792	.9192	1.0000	.1418	.7232	.6966					
.7493	-.3468	.5916	.8997									
.8353	-.1839	.6355	.8318									
.8791	-.0873	.6603	.7936									
.9212	-.0043	.6832	.7584									
1.0000	.1418	.7232	.6966									

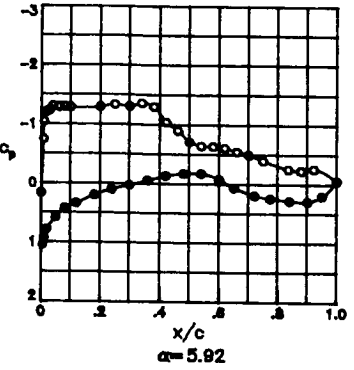
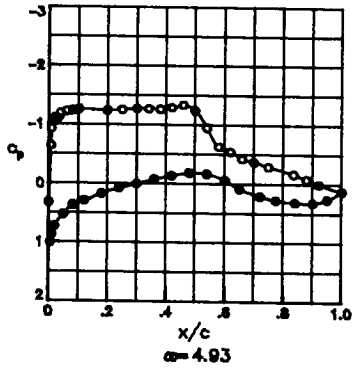
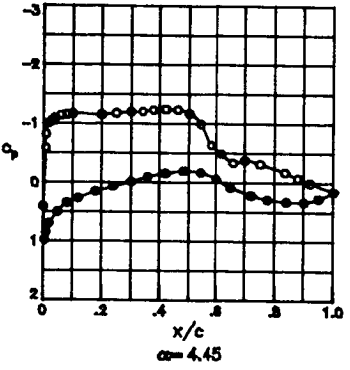
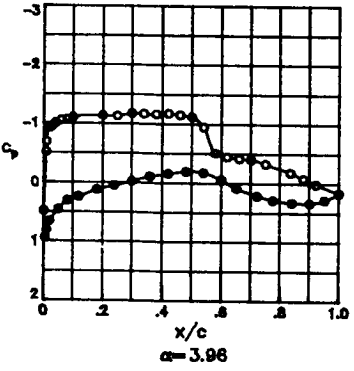
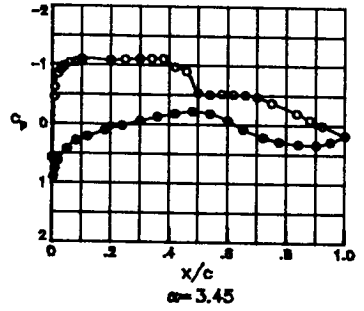
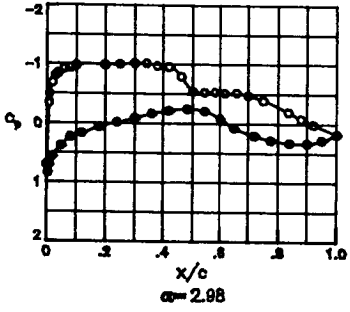
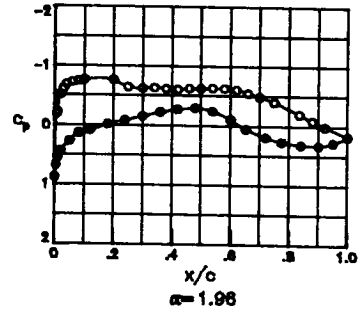
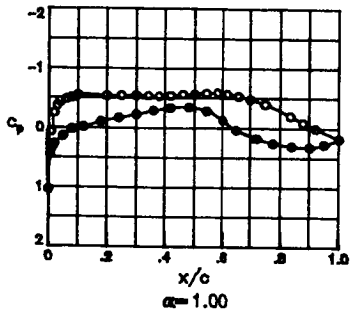
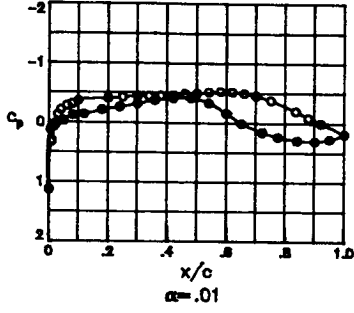
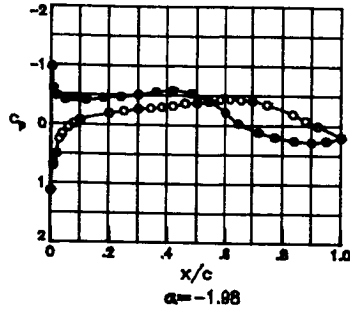
TEST 122
RUN 30
MACH .765
R 7.7×10^6



TFST	122	PT	17.7050	PSI		CN	.8920	CD1	.02509	CDCOR1	.02406
RUN	30	TT	130.2010	K		CM	-.1052	CD2	.02461	CDCOR2	.02365
POINT	9	RC	7.7955	MILLION		CC	-.0279	CD3	.02472	CDCOR3	.02378
		MACH	.7615					CD4	.02227	CDCOR4	.02146
		ALPHA	4.4070	DEG				CD5	.01956	CDCOR5	.01903

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _r L/PT	MLOC	X/C	CP	P _r L/PT	MLOC	X/C	Y/B ²	CP	P _r L/PT	MLOC
0.0000	.4065	.7937	.5842	0.0000	.4065	.7937	.5842	.0503	-.3375	-1.0423	.3945	1.2342
.0083	-.5727	.5233	1.0085	.0052	-.9748	.9487	.2726	.3957	-.3375	-1.1839	.3579	1.3066
.0697	-.9033	.4266	1.1742	.0098	-.8464	.9157	.3570	.5008	-.3375	-1.1500	.3614	1.2995
.0263	-1.0548	.3923	1.2384	.0200	.6955	.8730	.4449	.6048	-.3375	-.4941	.5449	.9735
.0300	-1.0690	.3842	1.2540	.0500	.4931	.8171	.5451	.7003	-.3375	-.3893	.5783	.9206
.0400	-1.1183	.3711	1.2799	.0813	.3504	.7772	.6112					
.0608	-1.1330	.3640	1.2902	.1199	.2615	.7523	.6511					
.0800	-1.1440	.3622	1.2978	.1796	.1486	.7204	.7010					
.1000	-1.1619	.3563	1.3100	.2397	.0719	.7015	.7303					
.1997	-1.1927	.3528	1.3171	.2995	-.0075	.6798	.7637					
.2500	-1.2075	.3489	1.3251	.3588	-.0825	.6592	.7953					
.2994	-1.2263	.3456	1.3322	.4193	-.1414	.6441	.8186					
.3402	-1.2286	.3445	1.3345	.4793	-.1878	.6310	.8387					
.3795	-1.2127	.3390	1.3461	.5394	-.1743	.6288	.8422					
.4201	-1.2668	.3395	1.3450	.5994	-.0543	.6708	.7776					
.4598	-1.2380	.3409	1.3422	.6507	.0909	.7072	.7215					
.4996	-1.1557	.3605	1.3014	.7203	.2165	.7404	.6698					
.5397	-.8848	.4322	1.1641	.7743	.2813	.7567	.6440					
.5795	-.5630	.5278	1.0012	.8394	.3257	.7724	.6189					
.6197	-.4777	.5504	.9647	.8996	.3254	.7719	.6198					
.6598	-.4092	.5678	.9371	.9492	.2780	.7579	.6422					
.6997	-.3661	.5769	.9229	1.0000	.1358	.7182	.7044					
.7493	-.3375	.5916	.8998									
.8353	-.1764	.6329	.8358									
.8791	-.0818	.6584	.7966									
.9212	-.0018	.6817	.7608									
1.0000	.1358	.7182	.7644									

TEST 122
 RUN 38
 MACH .765
 R 14.0×10^6

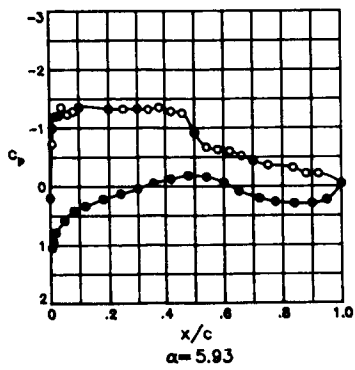
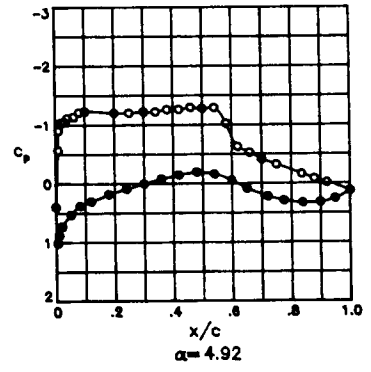
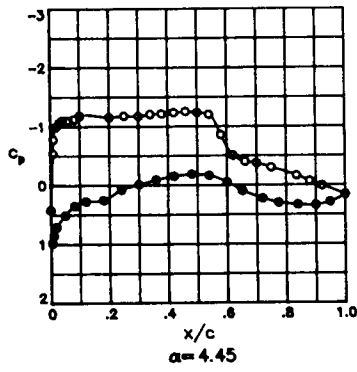
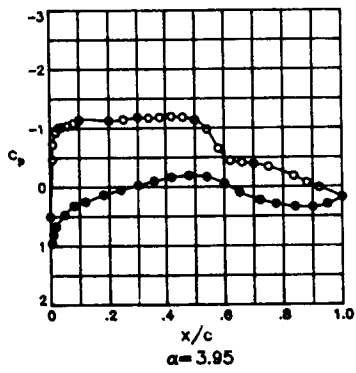
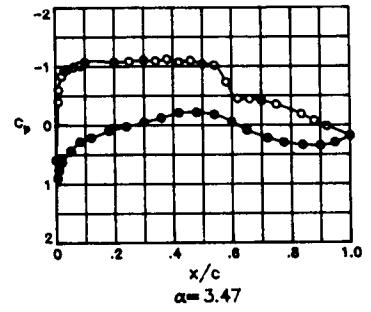
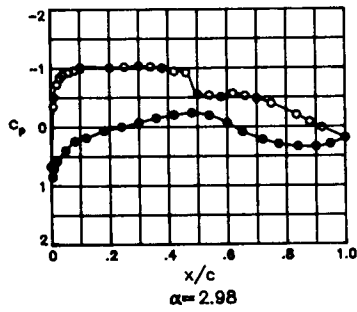
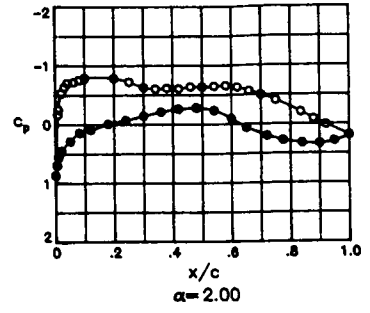
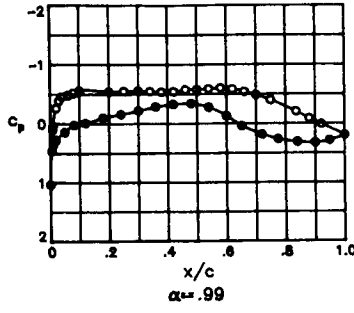
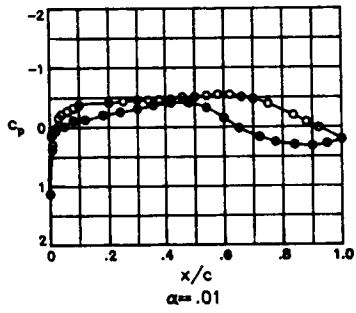
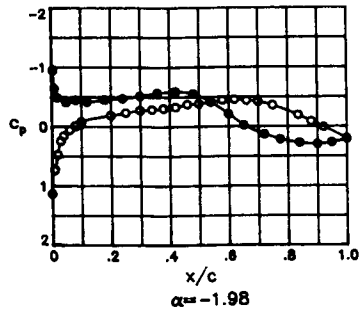


TEST	122	PT	23.2068	PSI		CN	.9570
RUN	36	TT	104.9283	K		CM	-.1133
POINT	10	RC	14.0350	MILLION		CC	-.0234
		MACH	.7619				
		ALPHA	5.9226	DEG			

CD1	.05257	CDCDR1	.05228
CD2	.05936	CDCDR2	.05901
CD3	.08034	CDCDR3	.07985
CD4	.04381	CDCDR4	.04368
CD5	.04069	CDCDR5	.04074

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC	X/C	Y/B/2	CP	P _L /PT	MLOC
0.0000	-.1703	.7330	.6821	0.0000	.1703	-.7330	-.6821	.0500	-.3375	-1.1087	.3742	1.2745
.0093	-.7447	.4838	1.0749	.0052	1.0477	-.9703	-.2083	.3957	-.3375	-1.3430	.3178	1.3932
.0097	-1.0422	.3909	1.2419	.0098	.9286	-.9382	-.3038	.5008	-.3375	-.7894	.4614	1.1134
.0203	-1.2134	.3483	1.3274	.0200	.7849	-.8986	-.3944	.6048	-.3375	-.5805	.5262	1.0047
.0300	-1.2456	.3394	1.3462	.0500	.5793	-.8413	-.5037	.7003	-.3375	-.4355	.5613	.9483
.0400	-1.3117	.3191	1.3904	.0813	.4220	-.7986	-.5768					
.0608	-1.2951	.3257	1.3758	.1199	.3386	-.7756	-.6145					
.0800	-1.2591	.3246	1.3781	.1796	.2032	-.7334	-.6815					
.1000	-1.2473	.3168	1.3955	.2397	.1060	-.7068	-.7229					
.1997	-1.2938	.3202	1.3879	.2995	.0257	-.6869	-.7543					
.2500	-1.3323	.3159	1.3975	.3588	-.0554	-.6574	-.7836					
.2994	-1.3078	.3117	1.4071	.4193	-.1374	-.6386	-.8279					
.3442	-1.3560	.3142	1.4015	.4793	-.1747	-.6373	-.8299					
.3795	-1.3002	.3158	1.3978	.5394	-.1775	-.6285	-.8434					
.4201	-1.0445	.3893	1.2452	.5994	-.0806	-.6569	-.7998					
.4598	-.9048	.4286	1.1715	.6507	.0694	-.6989	-.7351					
.4996	-.7712	.4815	1.0789	.7203	.1952	-.7333	-.6817					
.5397	-.6422	.4939	1.0579	.7743	.2453	-.7434	-.6659					
.5795	-.6358	.5028	1.0431	.8394	.2844	-.7582	-.6423					
.6197	-.6087	.5150	1.0230	.8996	.2958	-.7640	-.6331					
.6598	-.5488	.5284	1.0011	.9492	.2011	-.7360	-.6774					
.6997	-.4950	.5413	.9802	1.0000	-.0586	-.6618	-.7922					
.7493	-.3983	.5728	.9301									
.8393	-.2562	.6081	.8750									
.8791	-.2333	.6161	.8626									
.9212	-.2612	.6044	.8807									
1.0000	-.0586	.6618	.7922									

TEST 122
 RUN 43
 MACH .765
 R 30.0×10^6



TEST 122 PT 54.2551 PSI CN -.0023
RUN 43 TT 110.7275 K CM -.0984
POINT 1 RC 30.3940 MILLION CC .0047
MACH .7829
ALPHA -1.9760 DEG

CD1 .00669 CDCOR1 .00664
CD2 .00660 CDCOR2 .00653
CD3 .01697 CDCOR3 .01688
CD4 .00650 CDCOR4 .00644
CD5 .00639 CDCOR5 .00636

Table with 4 columns: UPPER SURFACE (X/C, CP, P/L/P, MLOC) and LOWER SURFACE (X/C, CP, P/L/P, MLOC). Values range from 0.0000 to 1.0000 and -0.0083 to .7405.

Table with 5 columns: SPANWISE (X/C, Y/B/2, CP, P/L/P, MLOC). Values range from .0500 to .7003 and -.3375 to .9686.

TEST 122 PT 54.2566 PSI CN .2832
RUN 43 TT 110.9533 K CM -.1033
POINT 2 RC 30.2440 MILLION CC .0059
MACH .7605
ALPHA .0136 DEG

CD1 .00652 CDCOR1 .00649
CD2 .00645 CDCOR2 .00640
CD3 .01677 CDCOR3 .01673
CD4 .00647 CDCOR4 .00645
CD5 .00630 CDCOR5 .00631

Table with 4 columns: UPPER SURFACE (X/C, CP, P/L/P, MLOC) and LOWER SURFACE (X/C, CP, P/L/P, MLOC). Values range from 0.0000 to 1.0000 and -0.0083 to .7550.

Table with 5 columns: SPANWISE (X/C, Y/B/2, CP, P/L/P, MLOC). Values range from .0500 to .7003 and -.3375 to .9603.

TEST 122 PT 54.2576 PSI CN .4198
RUN 43 TT 111.1184 K CM -.1044
POINT 3 RC 30.1280 MILLION CC .0010
MACH .7585
ALPHA .9886 DEG

CD1 .00664 CDCOR1 .00658
CD2 .00662 CDCOR2 .00651
CD3 .01691 CDCOR3 .01681
CD4 .00658 CDCOR4 .00652
CD5 .00643 CDCOR5 .00641

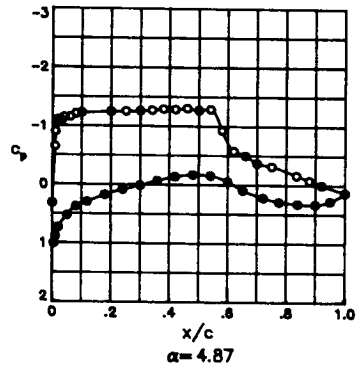
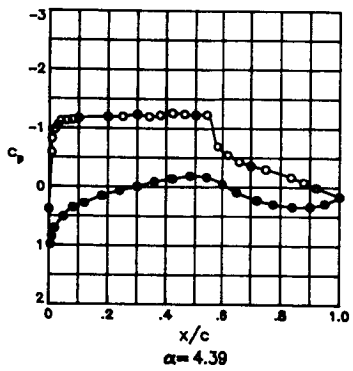
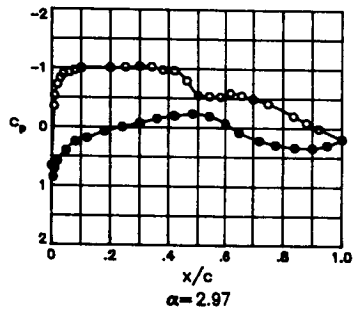
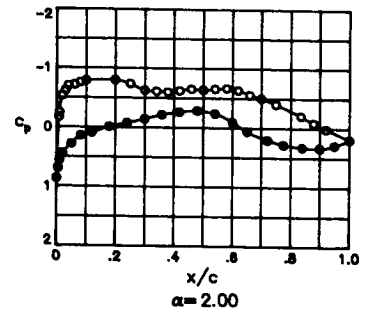
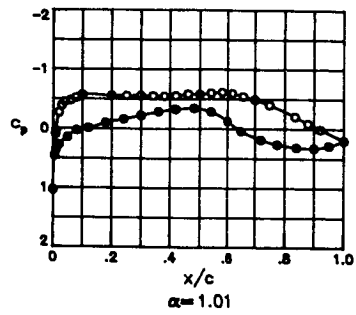
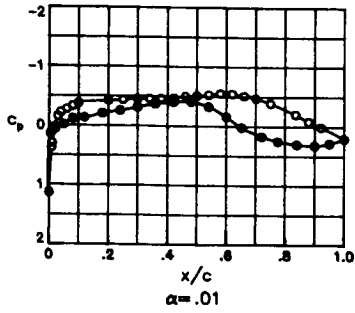
Table with 4 columns: UPPER SURFACE (X/C, CP, P/L/P, MLOC) and LOWER SURFACE (X/C, CP, P/L/P, MLOC). Values range from 0.0000 to 1.0000 and -0.0083 to .7534.

Table with 5 columns: SPANWISE (X/C, Y/B/2, CP, P/L/P, MLOC). Values range from .0500 to .7003 and -.3375 to .9658.

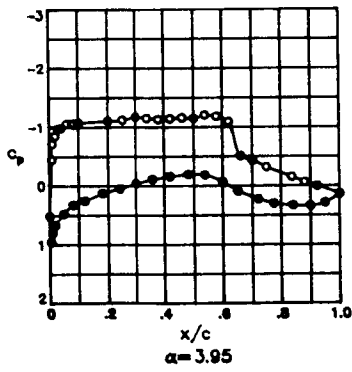
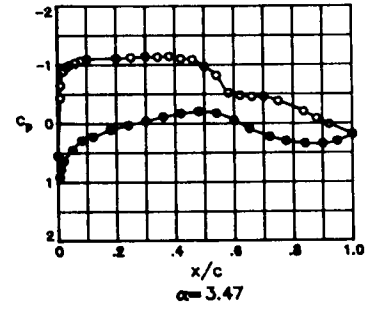
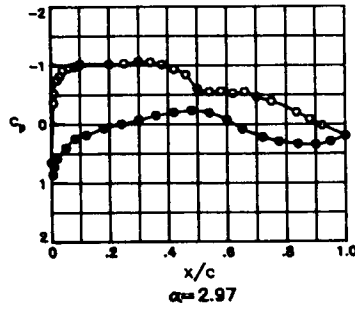
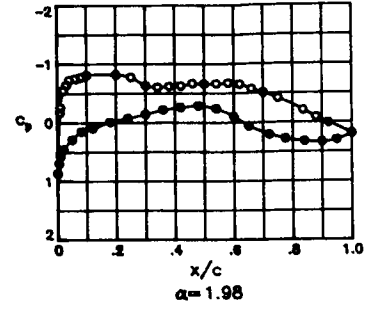
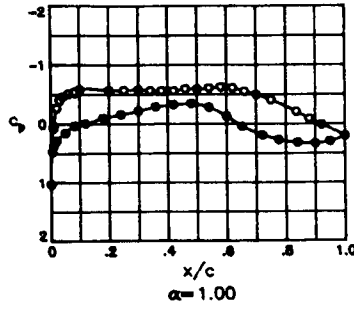
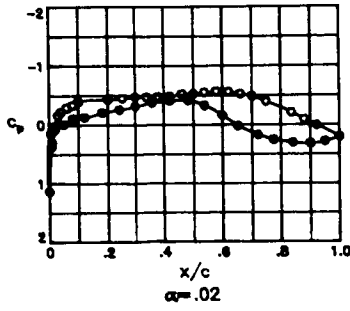
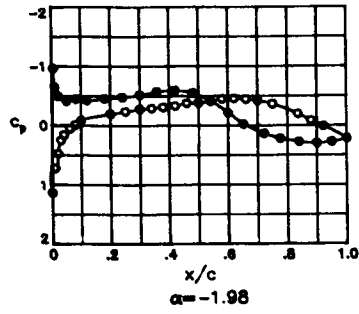
TEST	122	PT	55.3676	PSI		CN	.9985	CD1	.05935	CDCDR1	.05899
RUN	43	TT	113.0504	K		CM	-.1196	CD2	.06326	CDCDR2	.06283
POINT	10	RC	29.9360	MILLION		CC	-.0262	CD3	.11895	CDCDR3	.11848
		MACH	.7604					CD4	.04240	CDCDR4	.04207
		ALPHA	5.9299	DEG				CD5	.03787	CDCDR5	.03723

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/Z	CP	P/L/PT	MLOC
0.0000	.1962	.7370	.6765	0.0000	.1962	.7370	.6765	.0500	-.3375	-1.1220	.3770	1.2705
.0083	-.7271	.4633	1.0771	.0052	1.0475	.9695	.2114	.3957	-.3375	-1.3177	.3206	1.3883
.0097	-.9952	.3994	1.2271	.0098	.9331	.9394	.3008	.5008	-.3375	-.9671	.4211	1.1865
.0203	-1.1992	.3526	1.3197	.0200	.7833	.8967	.3986	.6048	-.3375	-.5915	.5172	1.0204
.0300	-1.2117	.3431	1.3396	.0500	.5884	.8458	.4961	.7003	-.3375	-.4230	.5688	.9374
.0400	-1.3526	.3147	1.4015	.0813	.4165	.7946	.5839					
.0608	-1.2303	.3371	1.3523	.1199	.3386	.7745	.6188					
.0800	-1.2426	.3264	1.3755	.1796	.2169	.7444	.6649					
.1000	-1.3563	.3141	1.4029	.2397	.1278	.7214	.7010					
.1997	-1.3263	.3213	1.3875	.2995	.0404	.6954	.7412					
.2500	-1.3249	.3162	1.3938	.3598	-.0519	.6684	.7828					
.2994	-1.3286	.3140	1.4031	.4193	-.1243	.6468	.8162					
.3402	-1.3148	.3160	1.3987	.4793	-.1831	.6294	.8429					
.3795	-1.3522	.3125	1.4066	.5394	-.1602	.6397	.8271					
.4201	-1.2067	.3329	1.3614	.5994	-.0666	.6667	.7855					
.4598	-1.2540	.3432	1.3395	.6507	.0826	.7081	.7215					
.4996	-.9155	.4337	1.1637	.7203	.2029	.7400	.6718					
.5397	-.6698	.4965	1.0548	.7743	.2648	.7545	.6489					
.5795	-.6308	.5081	1.0354	.8394	.2924	.7626	.6359					
.6197	-.6029	.5176	1.0199	.8996	.2897	.7628	.6356					
.6598	-.5227	.5445	.9762	.9492	.2282	.7489	.6578					
.6997	-.4349	.5699	.9358	1.0000	-.0461	.6722	.7770					
.7493	-.3638	.5829	.9153									
.8353	-.3198	.5922	.9008									
.8791	-.2210	.6221	.8542									
.9212	-.2186	.6203	.8569									
1.0000	-.0461	.6722	.7770									

TEST 122
 RUN 43
 MACH .765
 R 30.0×10^6



TEST 122
 RUN 52
 MACH .765
 R 45.0×10^6



TEST 122 PT 76.5314 PSI CM .9046
 RUN 52 TT 106.9794 K CM -.1293
 POINT 7 PC 49.2400 MILLION CC -.0205
 MACH .7701
 ALPHA 3.9500 DEG

CD1 .02187 CDCOR1 .01990
 CD2 .02113 CDCOR2 .02037
 CD3 .02203 CDCOR3 .02113
 CD4 .02326 CDCOR4 .02226
 CD5 .02263 CDCOR5 .02178

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/8/2	CP	P/L/PT	MLOC
0.0000	.5183	.8234	.5365	0.0000	.5183	.8234	.5365	.0500	-.3375	-.7842	.4599	1.1192
.0083	-.4445	.5567	.9585	.0052	.9594	.9469	.2813	.3957	-.3375	-1.1007	.3690	1.2884
.0097	-.7156	.4878	1.0712	.0098	.8015	.8997	.3930	.5008	-.3375	-1.1269	.3641	1.2983
.0203	-.8505	.4367	1.1605	.0200	.6694	.8654	.4610	.6048	-.3375	-1.1835	.3521	1.3226
.0300	-.9738	.4106	1.2079	.0500	.4755	.8103	.5989	.7003	-.3375	-.4241	.5557	.9601
.0400	-.9803	.4051	1.2183	.0813	.3252	.7702	.6250					
.0608	-1.0565	.3878	1.2512	.1199	.2557	.7493	.6584					
.0800	-1.0569	.3843	1.2581	.1796	.1278	.7101	.7200					
.1000	-1.0681	.3741	1.2782	.2397	.0486	.6893	.7522					
.1997	-1.1042	.3707	1.2850	.2995	-.0370	.6677	.7856					
.2500	-1.1184	.3667	1.2930	.3588	-.1110	.6470	.8174					
.2994	-1.1685	.3602	1.3062	.4193	-.1612	.6376	.8319					
.3402	-1.1515	.3605	1.3055	.4793	-.1941	.6258	.8501					
.3795	-1.1305	.3563	1.3141	.5394	-.1831	.6226	.8552					
.4201	-1.1419	.3550	1.3188	.5994	-.0649	.6569	.8022					
.4598	-1.1565	.3571	1.3124	.6507	.0942	.7047	.7283					
.4996	-1.1454	.3559	1.3149	.7203	.2303	.7405	.6723					
.5397	-1.2015	.3496	1.3279	.7743	.3047	.7653	.6329					
.5795	-1.1802	.3475	1.3324	.8394	.3368	.7709	.6239					
.6197	-1.0872	.3734	1.2796	.8996	.3459	.7734	.6199					
.6598	-.5054	.5403	.9848	.9492	.2929	.7612	.6394					
.6997	-.4254	.5543	.9623	1.0000	.1419	.7150	.7124					
.7493	-.3152	.5935	.9004									
.8353	-.1491	.6331	.8389									
.8791	-.0589	.6604	.7967									
.9212	.0061	.6796	.7672									
1.0000	.1419	.7150	.7124									

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Appendix F

Pressure Data for $M = 0.78$; $R = 4.4 \times 10^6$, 7.7×10^6 , 14.0×10^6 , 30.0×10^6 , and 45.0×10^6 ; and Free Transition

The pressure measurements made on the Boeing BAC I airfoil are presented in coefficient form in graphs and tables in this appendix. The data are for a Mach number of 0.78; Reynolds numbers of 4.4×10^6 , 7.7×10^6 , 14.0×10^6 , 30.0×10^6 , and 45.0×10^6 ; and free transition. The pressure data from the upper surface of the airfoil are plotted as open symbols, and the data from the lower surface are plotted as solid symbols.

TEST 122 PT 17.6842 PSI CN .7962 CD1 .01609 CDCOR1 .01480
RUN 19 TT 195.0954 K CM -.1188 CD2 .01560 CDCOR2 .01462
POINT 7 RC 4.4393 MILLION CC -.0156 CD3 .01727 CDCOR3 .01608
MACH 7.8488 CD4 .01720 CDCOR4 .01606
ALPHA 3.4565 DEG CD5 .01569 CDCOR5 .01472
UPPER SURFACE LOWER SURFACE SPANWISE
X/C CP P/L/PT MLOC X/C CP P/L/PT MLOC X/C Y/B/2 CP P/L/PT MLOC
0.0000 .7082 .8684 .4533 0.0000 .7082 .8684 .4533 .0500 -.3375 -.6803 .4719 1.0939
.0083 -.3782 .9552 .9567 .0052 .9023 .9253 .3349 .3957 -.3375 -1.0013 .3776 1.2665
.0097 -.5766 .9012 1.0443 .0098 .7648 .8862 .4189 .5008 -.3375 -1.0266 .3701 1.2015
.0203 -.7402 .4555 1.1222 .0200 .6166 .8423 .5011 .6048 -.3375 -.8975 .4079 1.2084
.0300 -.7956 .4396 1.1573 .0500 .4068 .7814 .6040 .7003 -.3375 -.3899 .5520 .9617
.0400 -.8572 .4168 1.1917 .0813 .2894 .7502 .6539
.0608 -.9438 .3975 1.2280 .1199 .1979 .7228 .6968
.0800 -.9595 .3906 1.2413 .1796 .0901 .6940 .7415
.1000 -.9853 .3869 1.2484 .2397 .0036 .6657 .7833
.1997 -1.0065 .3777 1.2663 .2995 -.0789 .6438 .8186
.2500 -1.0279 .3723 1.2760 .3588 -.1582 .6218 .8524
.2994 -1.0421 .3661 1.2894 .4193 -.2263 .6005 .8853
.3402 -1.0506 .3641 1.2936 .4793 -.2630 .5903 .9013
.3795 -1.0692 .3614 1.2990 .5394 -.2282 .6021 .8830
.4201 -1.0539 .3670 1.2877 .5994 -.0908 .6421 .8211
.4598 -1.0550 .3628 1.2961 .6507 .0823 .6895 .7484
.4996 -1.0618 .3623 1.2971 .7203 .2231 .7306 .6847
.5397 -1.0537 .3632 1.2954 .7743 .2999 .7519 .6512
.5795 -.9809 .3856 1.2509 .8394 .3416 .7646 .6311
.6197 -.7197 .4619 1.1113 .8996 .3464 .7667 .6278
.6598 -.4748 .5276 1.0610 .9492 .3066 .7528 .6498
.6997 -.3876 .5546 .9576 1.0000 .1311 .7029 .7277
.7493 -.3431 .5661 .9393
.8353 -.1658 .6183 .8579
.8791 -.0762 .6429 .8200
.9212 .0048 .6662 .7842
1.0000 .1311 .7029 .7277

TEST 122 PT 17.6544 PSI CN .8449 CD1 .02015 CDCOR1 .01907
RUN 19 TT 195.0558 K CM -.1166 CD2 .01915 CDCOR2 .01831
POINT 8 RC 4.4102 MILLION CC -.0198 CD3 .02065 CDCOR3 .01977
MACH 7.7773 CD4 .01993 CDCOR4 .01889
ALPHA 3.9391 DEG CD5 .01700 CDCOR5 .01631
UPPER SURFACE LOWER SURFACE SPANWISE
X/C CP P/L/PT MLOC X/C CP P/L/PT MLOC X/C Y/B/2 CP P/L/PT MLOC
0.0000 .5956 .8391 .5067 0.0000 .5956 .8391 .5067 .0500 -.3375 -.7381 .4622 1.1106
.0083 -.4730 .5351 .9889 .0052 .9579 .9426 .2915 .3957 -.3375 -1.0805 .3638 1.2942
.0097 -.7190 .4673 1.1022 .0098 .8127 .9012 .3882 .5008 -.3375 -1.1028 .3616 1.2986
.0203 -.8520 .4284 1.1705 .0200 .6668 .8605 .4681 .6048 -.3375 -.8024 .4380 1.1532
.0300 -.9191 .4115 1.2015 .0500 .4458 .7964 .5794 .7003 -.3375 -.3747 .5651 .9409
.0400 -.9505 .3990 1.2291 .0813 .3215 .7625 .6343
.0608 -1.0258 .3608 1.2602 .1199 .2315 .7370 .6747
.0800 -1.0458 .3750 1.2716 .1796 .1175 .7047 .7249
.1000 -1.0612 .3708 1.2781 .2397 .0341 .6796 .7635
.1997 -1.0763 .3646 1.2925 .2995 -.0529 .6553 .8009
.2500 -1.0921 .3589 1.3040 .3588 -.1359 .6310 .8382
.2994 -1.1025 .3515 1.3193 .4193 -.2031 .6090 .8722
.3402 -1.1209 .3515 1.3194 .4793 -.2421 .6013 .8842
.3795 -1.1395 .3476 1.3274 .5394 -.2136 .6103 .8702
.4201 -1.1423 .3474 1.3280 .5994 -.0822 .6479 .8123
.4598 -1.1291 .3503 1.3220 .6507 .0857 .6950 .7398
.4996 -1.1355 .3479 1.3271 .7203 .2239 .7339 .6795
.5397 -1.1334 .3473 1.3282 .7743 .2992 .7549 .6466
.5795 -.8629 .4275 1.1721 .8394 .3423 .7687 .6245
.6197 -.6015 .4567 1.0518 .8996 .3431 .7664 .6283
.6598 -.4449 .5420 .9777 .9492 .3045 .7557 .6453
.6997 -.3820 .5591 .9503 1.0000 .1277 .7079 .7199
.7493 -.3402 .5756 .9243
.8353 -.1710 .6216 .8527
.8791 -.0837 .6482 .8117
.9212 -.0020 .6691 .7797
1.0000 .1277 .7079 .7199

TEST 122 PT 17.6944 PSI CN .8894 CD1 .02694 CDCOR1 .02588
RUN 19 TT 195.4065 K CM -.1199 CD2 .02613 CDCOR2 .02503
POINT 9 RC 4.4122 MILLION CC -.0210 CD3 .02976 CDCOR3 .02806
MACH 7.785 CD4 .02572 CDCOR4 .02453
ALPHA 4.4176 DEG CD5 .02101 CDCOR5 .02016
UPPER SURFACE LOWER SURFACE SPANWISE
X/C CP P/L/PT MLOC X/C CP P/L/PT MLOC X/C Y/B/2 CP P/L/PT MLOC
0.0000 .5303 .8210 .5381 0.0000 .5303 .8210 .5381 .0500 -.3375 -.7746 .4515 1.1292
.0083 -.5270 .5267 1.0122 .0052 .9955 .9528 .2636 .3957 -.3375 -1.1187 .3493 1.3239
.0097 -.7968 .4423 1.1455 .0098 .8481 .9104 .3685 .5008 -.3375 -1.1473 .3455 1.3318
.0203 -.9024 .4105 1.2035 .0200 .6995 .8683 .4535 .6048 -.3375 -.7163 .4680 1.1005
.0300 -.9619 .3947 1.2334 .0500 .4418 .8072 .5615
.0400 -1.0143 .3821 1.2576 .0813 .3524 .7689 .6241
.0608 -1.0341 .3675 1.2667 .1199 .2383 .7455 .6613
.0800 -1.0974 .3625 1.2968 .1796 .1435 .7135 .7113
.1000 -1.1177 .3576 1.3068 .2397 .0530 .6864 .7531
.1997 -1.1061 .3527 1.3169 .2995 -.0353 .6584 .7962
.2500 -1.1397 .3471 1.3286 .3588 -.1190 .6368 .8293
.2994 -1.1654 .3407 1.3417 .4193 -.1826 .6195 .8560
.3402 -1.1633 .3388 1.3461 .4793 -.2297 .6044 .8793
.3795 -1.1830 .3342 1.3559 .5394 -.2044 .6122 .8672
.4201 -1.1937 .3311 1.3626 .5994 -.0839 .6444 .8145
.4598 -1.1842 .3325 1.3597 .6507 .0848 .6937 .7410
.4996 -1.1772 .3349 1.3544 .7203 .2268 .7343 .6789
.5397 -1.1864 .3352 1.3538 .7743 .2996 .7563 .6443
.5795 -.9111 .4163 1.2029 .8394 .3398 .7665 .6280
.6197 -.5583 .5038 1.0401 .8996 .3411 .7675 .6264
.6598 -.4493 .5296 .9977 .9492 .2926 .7542 .6477
.6997 -.3930 .5603 .9484 1.0000 .1012 .6991 .7335
.7493 -.3501 .5713 .9311
.8353 -.1736 .6209 .8539
.8791 -.0918 .6436 .8188
.9212 -.0203 .6644 .7870
1.0000 .1012 .6991 .7335

TEST 122	PT	17.6903	PSI	CN	.9304	CD1	.03579	CDCOR1	.03455
RUN 19	TT	195.8500	K	CM	-1.230	CD2	.03492	CDCOR2	.03373
POINT 10	RC	4.3972	MILLION	CC	-0.0219	CD3	.03853	CDCOR3	.03728
	MACH	.7785				CD4	.03080	CDCOR4	.02923
	ALPHA	4.9171	DEG			CD5	.02277	CDCOR5	.02166

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _r L/PT	MLOC	X/C	CP	P _r L/PT	MLOC	X/C	Y/8/2	CP	P _r L/PT	MLOC
0.0000	.4637	.8008	.5722	0.0000	.4637	.8008	.5722	.0500	-.3375	-.8034	.4409	1.1481
.0093	-.4639	.5075	1.0338	.0093	1.0241	.9611	.2386	.3957	-.3375	-1.1850	.3365	1.3511
.0497	-.9023	.4131	1.1985	.0098	.8853	.9216	.3434	.5008	-.3375	-1.1952	.3324	1.3598
.0203	-.9840	.3897	1.2430	.0200	.7567	.8800	.4311	.6048	-.3375	-.6747	.4790	1.0817
.0300	-1.0352	.3775	1.2666	.0500	.5100	.8148	.5487	.7003	-.3375	-.4569	.5389	.9827
.0400	-1.0723	.3645	1.2927	.0813	.3792	.7788	.6082					
.0608	-1.1225	.3533	1.3156	.1199	.2822	.7497	.6547					
.0800	-1.1360	.3458	1.3312	.1796	.1655	.7152	.7087					
.1000	-1.1380	.3427	1.3378	.2397	.0709	.6898	.7479					
.1497	-1.1617	.3397	1.3441	.2995	-.0204	.6642	.7872					
.2500	-1.1803	.3335	1.3574	.3588	-.1042	.6399	.8246					
.2994	-1.2030	.3284	1.3687	.4193	-.1786	.6195	.8561					
.3402	-1.1977	.3262	1.3735	.4793	-.2265	.6035	.8808					
.3795	-1.2268	.3221	1.3824	.5394	-.2653	.6122	.8673					
.4201	-1.2429	.3188	1.3898	.5994	-.0834	.6475	.8329					
.4598	-1.2549	.3150	1.3984	.6507	.0820	.6942	.7411					
.4996	-1.2277	.3207	1.3855	.7203	.2159	.7312	.6837					
.5397	-1.2242	.3245	1.3771	.7743	.2931	.7544	.6473					
.5795	-.9332	.4675	1.2090	.8394	.3308	.7653	.6298					
.6197	-.6271	.4919	1.0598	.8996	.3333	.7649	.6306					
.6598	-.5161	.5241	1.0066	.9492	.2885	.7525	.6502					
.6997	-.4238	.5513	.9628	1.0000	.0864	.6952	.7396					
.7493	-.3645	.5655	.9403									
.8353	-.1968	.6137	.8650									
.8791	-.1063	.6387	.8264									
.9212	-.0327	.6611	.7919									
1.0000	.0864	.6952	.7396									

TEST 122	PT	17.6483	PSI	CN	1.0136	CD1	.05642	CDCOR1	.05501
RUN 19	TT	195.3279	K	CM	-.1395	CD2	.05789	CDCOR2	.05606
POINT 11	RC	4.4057	MILLION	CC	-.0191	CD3	.06349	CDCOR3	.06192
	MACH	.7791				CD4	.04851	CDCOR4	.04711
	ALPHA	5.8979	DEG			CD5	.03382	CDCOR5	.03279

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _r L/PT	MLOC	X/C	CP	P _r L/PT	MLOC	X/C	Y/8/2	CP	P _r L/PT	MLOC
0.0000	.3233	.7612	.6364	0.0000	.3233	.7612	.6364	.0500	-.3375	-.8839	.4182	1.1892
.0083	-.4673	.4790	1.0816	.0092	1.0777	.9763	.1854	.3957	-.3375	-1.2622	.3114	1.4065
.0497	-1.0430	.3727	1.2763	.0098	.9453	.9383	.3028	.5008	-.3375	-1.0668	.3643	1.2931
.0203	-1.1127	.3513	1.3197	.0200	.7967	.8961	.3989	.6048	-.3375	-.6998	.4719	1.0937
.0300	-1.1382	.3446	1.3338	.0500	.5691	.8321	.5190	.7003	-.3375	-.5482	.5147	1.0220
.0400	-1.2070	.3277	1.3700	.0813	.4321	.7926	.5858					
.0608	-1.2152	.3239	1.3787	.1199	.3277	.7618	.6354					
.0800	-1.2253	.3185	1.3905	.1796	.2046	.7275	.6895					
.1000	-1.2401	.3160	1.3960	.2397	.1081	.7002	.7319					
.1497	-1.2444	.3153	1.3975	.2995	.0149	.6738	.7725					
.2500	-1.2597	.3106	1.4084	.3588	-.0751	.6479	.8122					
.2994	-1.2801	.3067	1.4172	.4193	-.1537	.6267	.8449					
.3402	-1.2431	.3045	1.4223	.4793	-.2080	.6104	.8700					
.3795	-1.2924	.3013	1.4297	.5394	-.2048	.6110	.8691					
.4201	-1.2978	.2986	1.4361	.5994	-.0888	.6434	.8192					
.4598	-1.2984	.3003	1.4321	.6507	.0709	.6899	.7478					
.4996	-1.2059	.3272	1.3712	.7203	.2083	.7293	.6868					
.5397	-.9367	.4014	1.2205	.7743	.2849	.7499	.6544					
.5795	-.7195	.4615	1.1117	.8394	.3108	.7563	.6443					
.6197	-.6962	.4731	1.0918	.8996	.3097	.7586	.6406					
.6598	-.6453	.4850	1.0715	.9492	.2534	.7412	.6682					
.6997	-.5786	.5056	1.0370	1.0000	-.0509	.6549	.8016					
.7493	-.4862	.5315	.9947									
.8353	-.3308	.5742	.9265									
.8791	-.2185	.6052	.8781									
.9212	-.1483	.6104	.8701									
1.0000	-.0509	.6549	.8016									

TEST RUN POINT	122 27 2	PT TT RC	17.6772 131.6184 7.4225	PSI K MILLION	CN CM CC	-.0210 -.0917 .0047	CD1 CD2 CD3 CD4 CD5	.00847 .00834 .00835 .00837 .00783	CDCOR1 CDCOR2 CDCOR3 CDCOR4 CDCOR5	.00837 .00824 .00824 .00830 .00779
		MACH ALPHA	.7785 -2.610J	DEG						

TEST RUN POINT	122 27 3	PT TT PC	17.6757 131.5512 7.8376	PSI K MILLION	CN CM CC	-1.286 -.0960 .0071	CD1 CD2 CD3 CD4 CD5	.00827 .00822 .00806 .00829 .00744	CDCOR1 CDCOR2 CDCOR3 CDCOR4 CDCOR5	.00818 .00812 .00795 .00821 .00740
		MACH ALPHA	.7868 -.9900	DEG						

TEST RUN POINT	122 27 4	PT TT RC	17.6760 131.6576 7.6194	PSI K MILLION	CN CM CC	-.2682 -.0983 .0058	CD1 CD2 CD3 CD4 CD5	.00825 .00815 .00811 .00808 .00736	CDCOR1 CDCOR2 CDCOR3 CDCOR4 CDCOR5	.00814 .00802 .00798 .00799 .00730
		MACH ALPHA	.7794 -.0014	DEG						

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TEST	122	PT	17.6541	PSI	CN	.R916	CD1	.04063	CDCDR1	.03953
RUN	27	TT	132.0304	K	CM	-.1090	CD2	.04115	CDCDR2	.03989
POINT	12	RC	7.7240	MILLION	CC	-.0226	CD3	.04001	CDCDR3	.03878
		MACH	.7803				CD4	.03999	CDCDR4	.02997
		ALPHA	4.9100	DEG			CD5	.02313	CDCDR5	.02240

UPPER SURFACE					LOWER SURFACE					SPANWISE				
X/C	CP	P,L/PT	MLOC		X/C	CP	P,L/PT	MLOC		X/C	Y/B/2	CP	P,L/PT	MLOC
0.0000	.4242	.7913	.5881		0.0000	.4242	.7913	.5881		.0503	-.3375	-.9500	.3968	1.2297
.0083	-.5676	.5102	1.0299		.0052	.9806	.9480	.2773		.3957	-.3375	-1.1558	.3402	1.3435
.0097	-.8498	.4247	1.1777		.0098	.8475	.9101	.3693		.5008	-.3375	-1.1437	.3374	1.3495
.0203	-.9755	.3895	1.2438		.0200	.6991	.9674	.4554		.6048	-.3375	-.5716	.5015	1.0444
.0300	-1.0084	.3791	1.2641		.0500	.4982	.9111	.5554		.7003	-.3375	-.4035	.5565	.9549
.0400	-1.0721	.3637	1.2947		.0813	.3541	.7668	.6279						
.0608	-1.0599	.3598	1.3027		.1199	.2704	.7478	.6581						
.0800	-1.1225	.3530	1.3167		.1796	.1519	.7122	.7136						
.1600	-1.1255	.3482	1.3267		.2397	.0724	.6926	.7440						
.1997	-1.1565	.3406	1.3427		.2995	-.0178	.6646	.7871						
.2500	-1.1508	.3374	1.3497		.3588	-.1067	.6364	.8304						
.2994	-1.1833	.3317	1.3618		.4193	-.1712	.6202	.8554						
.3402	-1.1657	.3291	1.3674		.4793	-.2324	.5979	.8900						
.3795	-1.1964	.3252	1.3760		.5394	-.1988	.6105	.8703						
.4201	-1.2234	.3221	1.3828		.5994	-.0807	.6470	.8141						
.4598	-1.2312	.3192	1.3893		.6507	.0689	.6891	.7493						
.4996	-1.1884	.3308	1.3637		.7203	.1936	.7244	.6948						
.5397	-.9824	.3878	1.2471		.7743	.2616	.7429	.6658						
.5795	-.6714	.4810	1.0787		.8394	.3043	.7573	.6428						
.6197	-.3761	.5036	1.0408		.8996	.2992	.7536	.6489						
.6598	-.4992	.5300	.9976		.9492	.2313	.7369	.6752						
.6997	-.4519	.5373	.9858		1.0000	.0611	.6870	.7526						
.7493	-.3527	.5688	.9355											
.8353	-.1917	.6165	.8610											
.8791	-.1185	.6344	.8335											
.9212	-.0307	.6616	.7916											
1.0000	.0611	.6870	.7526											

TEST	122	PT	21.1723	PSI	CN	.7680	CD1	.01391	CDCOR1	.01326
RUN	34	TT	100.1255	K	CM	-.1052	CD2	.01479	CDCOR2	.01408
POINT	7	RC	13.8760	MILLION	CC	-.0189	CD3	.01941	CDCOR3	.01474
		MACH	.7765				CD4	.01549	CDCOR4	.01507
		ALPHA	3.4600	DEG			CD5	.01532	CDCOR5	.01503

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/2	CP	P/L/PT	MLOC
0.0000	.6253	.8507	.4869	0.0000	.6253	.8507	.4869	.0500	-.3375	-.7304	.4719	1.0954
.0083	-.3923	.5655	.9417	.0052	.8758	.9196	.3487	.3957	-.3375	-1.0144	.3866	1.2504
.0097	-.5209	.5246	1.0074	.0098	.7406	.8817	.4286	.5008	-.3375	-1.0405	.3840	1.2559
.0203	-.7267	.4675	1.1029	.0200	.6004	.8422	.5022	.6048	-.3375	-.9899	.5038	1.0415
.0300	-.8276	.4392	1.1525	.0500	.4160	.7908	.5897	.7003	-.3375	-.3784	.5656	.9416
.0400	-.8947	.4219	1.1839	.0813	.2665	.7489	.6572					
.0608	-.9432	.4084	1.2087	.1199	.1987	.7278	.6904					
.0800	-.9883	.4003	1.2241	.1796	.0939	.6987	.7355					
.1000	-.9915	.3919	1.2400	.2397	.0100	.6755	.7712					
.1997	-1.0315	.3847	1.2541	.2995	-.0646	.6564	.8007					
.2500	-1.0337	.3805	1.2623	.3588	-.1461	.6311	.8395					
.2994	-1.0614	.3746	1.2739	.4193	-.2009	.6170	.8614					
.3402	-1.0692	.3742	1.2746	.4793	-.2341	.6088	.8739					
.3795	-1.0551	.3704	1.2822	.5394	-.2019	.6127	.8678					
.4231	-1.0336	.3781	1.2670	.5994	-.0731	.6503	.8099					
.4598	-1.0558	.3765	1.2701	.6507	-.0782	.6598	.7400					
.4996	-1.0355	.3806	1.2621	.7203	.2126	.7328	.6826					
.5397	-.8848	.4286	1.1753	.7743	.2823	.7542	.6449					
.5795	-.7045	.4779	1.0850	.8394	.3277	.7673	.6279					
.6197	-.4476	.5375	.9864	.8996	.3344	.7684	.6260					
.6598	-.4237	.5547	.9588	.9492	.2781	.7522	.6520					
.6997	-.3759	.5621	.9472	1.0000	.1595	.7165	.7080					
.7493	-.3293	.5610	.9174									
.8353	-.1662	.6232	.8518									
.8791	-.0720	.6536	.8049									
.9212	.0156	.6730	.7751									
1.0000	.1595	.7165	.7080									

TEST	122	PT	21.1918	PSI	CN	-.9366	CD1	.02011	CDCOR1	.01903
RUN	34	TT	100.2763	K	CM	-.1147	CD2	.02136	CDCOR2	.02052
POINT	8	RC	13.8880	MILLION	CC	-.0191	CD3	.02153	CDCOR3	.02065
		MACH	.7784				CD4	.02314	CDCOR4	.02206
		ALPHA	4.0098	DEG			CD5	.02192	CDCOR5	.02083

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/2	CP	P/L/PT	MLOC
0.0000	.5787	.8330	.5184	0.0000	.5787	.8330	.5184	.0500	-.3375	-.8096	.4429	1.1460
.0083	-.4285	.5453	.9740	.0052	.9326	.9373	.3060	.3957	-.3375	-1.0818	.3676	1.2879
.0097	-.6438	.4968	1.0531	.0098	.7813	.8922	.4077	.5008	-.3375	-1.0825	.3580	1.3073
.0203	-.8080	.4410	1.1493	.0200	.6405	.8527	.4834	.6048	-.3375	-.9975	.3841	1.2553
.0300	-.9059	.4147	1.1971	.0500	.4501	.7993	.5758	.7003	-.3375	-.3898	.5588	.9524
.0400	-.9394	.4066	1.2123	.0813	.2941	.7548	.6478					
.0608	-.9988	.3875	1.2486	.1199	.2243	.7333	.6817					
.0800	-1.0091	.3836	1.2562	.1796	.1122	.7008	.7323					
.1000	-1.0353	.3738	1.2755	.2397	.0348	.6835	.7589					
.1997	-1.0770	.3676	1.2879	.2995	-.0505	.6578	.7985					
.2500	-1.0685	.3620	1.2991	.3598	-.1377	.6281	.8441					
.2994	-1.1143	.3574	1.3086	.4193	-.1899	.6186	.8588					
.3402	-1.1124	.3557	1.3121	.4793	-.2302	.6057	.8788					
.3795	-1.1315	.3518	1.3201	.5394	-.2017	.6148	.8647					
.4201	-1.1282	.3521	1.3195	.5994	-.0963	.6471	.8150					
.4598	-1.1124	.3522	1.3184	.6507	.0671	.6881	.7518					
.4996	-1.1291	.3529	1.3179	.7203	.2065	.7304	.6862					
.5397	-1.1380	.3468	1.3306	.7743	.2766	.7486	.6576					
.5795	-1.0462	.3744	1.2742	.8394	.3129	.7597	.6401					
.6197	-.6581	.4906	1.0636	.8996	.3289	.7674	.6277					
.6598	-.4715	.5355	.9897	.9492	.2674	.7456	.6624					
.6997	-.3826	.5629	.9458	1.0000	.1276	.7059	.7245					
.7493	-.3133	.5835	.9133									
.8353	-.1618	.6279	.8446									
.8791	-.0688	.6501	.8103									
.9212	-.0426	.6692	.7810									
1.0000	.1276	.7059	.7245									

TEST	122	PT	21.0745	PSI	CN	-.9062	CD1	.02979	CDCOR1	.02920
RUN	34	TT	99.7375	K	CM	-.1296	CD2	.03285	CDCOR2	.03211
POINT	9	RC	14.5960	MILLION	CC	-.0180	CD3	.04378	CDCOR3	.04301
		MACH	.7855				CD4	.04086	CDCOR4	.04041
		ALPHA	4.4400	DEG			CD5	.03625	CDCOR5	.03598

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/2	CP	P/L/PT	MLOC
0.0000	.4857	.8692	.5591	0.0000	.4857	.8692	.5591	.0500	-.3375	-.8048	.4345	1.1611
.0083	-.5032	.5296	.9992	.0052	.9627	.9440	.2885	.3957	-.3375	-1.0786	.3536	1.3166
.0097	-.6461	.4773	1.0861	.0098	.8237	.9054	.3800	.5008	-.3375	-1.1334	.3460	1.3323
.0203	-.8497	.4226	1.1826	.0200	.6767	.8627	.4644	.6048	-.3375	-.6909	.4643	1.1089
.0300	-.9613	.3983	1.2279	.0500	.4801	.8061	.5645	.7003	-.3375	-.4392	.5317	.9960
.0400	-.9845	.3895	1.2448	.0813	.3293	.7661	.6299					
.0608	-1.0418	.3740	1.2751	.1199	.2582	.7460	.6618					
.0800	-1.0691	.3718	1.2794	.1796	.1376	.7099	.7182					
.1000	-1.0969	.3594	1.3035	.2397	.0522	.6869	.7538					
.1997	-1.1228	.3550	1.3135	.2995	-.0267	.6647	.7878					
.2500	-1.1249	.3527	1.3182	.3598	-.1134	.6392	.8271					
.2994	-1.1773	.3467	1.3308	.4193	-.1700	.6286	.8434					
.3402	-1.1414	.3438	1.3368	.4793	-.2281	.6040	.8815					
.3795	-1.1769	.3404	1.3439	.5394	-.1956	.6175	.8606					
.4201	-1.1871	.3370	1.3513	.5994	-.0726	.6511	.8076					
.4598	-1.2395	.3384	1.3442	.6507	.0795	.6988	.7354					
.4996	-1.1703	.3400	1.3449	.7203	.2045	.7293	.6880					
.5397	-1.1413	.3321	1.3618	.7743	.2765	.7488	.6573					
.5795	-1.0832	.3635	1.2961	.8394	.3167	.7606	.6387					
.6197	-.6164	.4978	1.0515	.8996	.3212	.7628	.6351					
.6598	-.5396	.5249	1.0608	.9492	.2596	.7433	.6660					
.6997	-.4400	.5536	.9924	1.0000	.0114	.6671	.7843					
.7493	-.3463	.5484	.9692									
.8353	-.3044	.5700	.9344									
.8791	-.1798	.6139	.8663									
.9212	-.0308	.6585	.7975									
1.0000	.0414	.6671	.7843									

TEST 122 PT 21.5692 PSI CN .9057
 RUN 34 TT 100.0582 K CM -.1179
 POINT 10 RC 14.1730 MILLION CC -.0203
 MACH .7793
 ALPHA 4.9321 DEG

CD1 .03922 CDCOR1 .03850
 CD2 .03892 CDCOR2 .03808
 CD3 .05520 CDCOR3 .05440
 CD4 .04024 CDCOR4 .03965
 CD5 .03600 CDCOR5 .03552

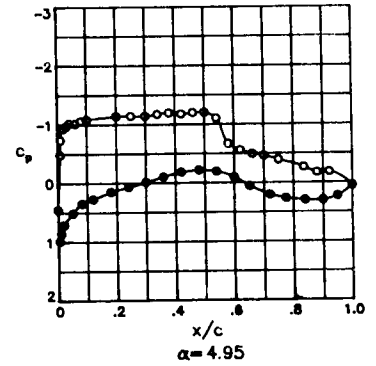
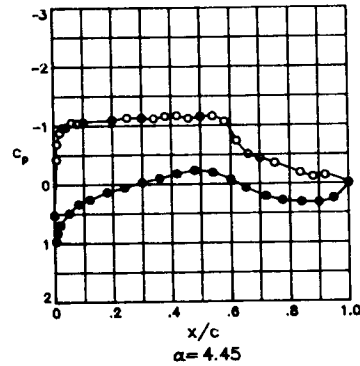
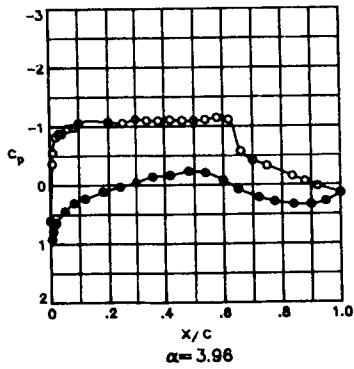
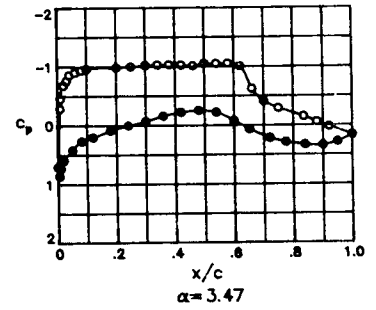
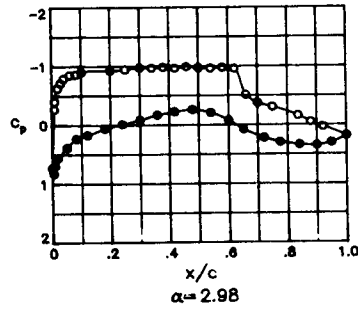
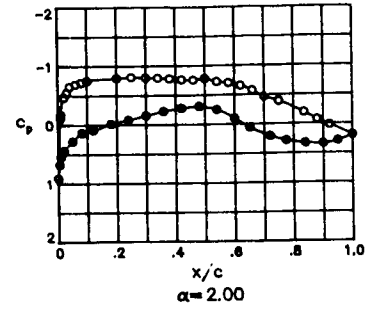
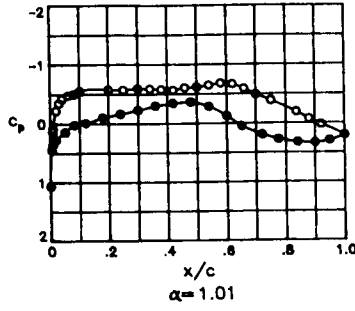
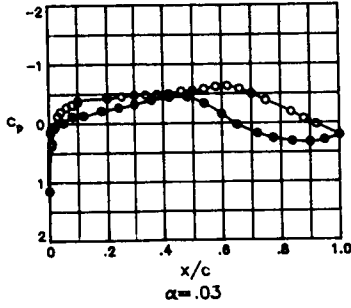
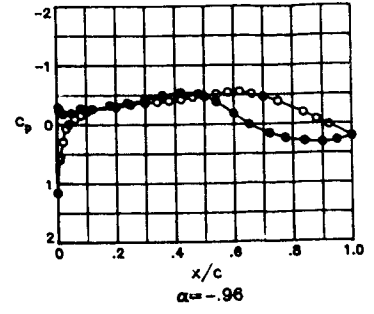
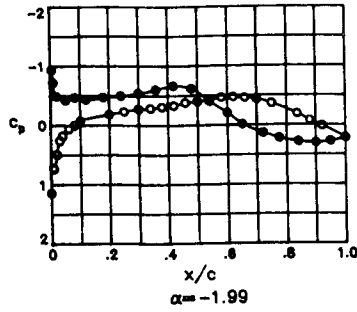
UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _s L/PT	MLOC	X/C	CP	P _s L/PT	MLOC	X/C	Y/B/2	CP	P _s L/PT	MLOC
0.0000	.4321	.7923	.5874	0.0000	.4321	.7923	.5874	.0500	-.3375	-.8860	.4181	1.1908
.0083	-.5480	.5134	1.0258	.0052	.9867	.9500	.2722	.3957	-.3375	-1.1542	.3428	1.3389
.0097	-.7822	.4458	1.1409	.0098	.8502	.9109	.3682	.5003	-.3375	-1.1760	.3357	1.3542
.0203	-.9334	.4021	1.2208	.0200	.7045	.8688	.4535	.6508	-.3375	-.6021	.4927	1.0600
.0300	-.9709	.3897	1.2445	.0500	.5121	.8158	.5481	.7003	-.3375	-.4336	.5506	.9655
.0400	-1.0613	.3692	1.2847	.0813	.3519	.7693	.6248					
.0608	-1.0866	.3597	1.3040	.1199	.2807	.7501	.6555					
.0800	-1.1153	.3537	1.3162	.1796	.1676	.7210	.7011					
.1000	-1.1587	.3477	1.3288	.2397	.0779	.6961	.7395					
.1997	-1.1673	.3445	1.3355	.2995	-.0078	.6712	.7780					
.2500	-1.1373	.3392	1.3468	.3588	-.1126	.6334	.8361					
.2994	-1.1697	.3325	1.3611	.4193	-.1709	.6182	.8594					
.3402	-1.2048	.3223	1.3614	.4793	-.2040	.6149	.8646					
.3795	-1.2092	.3293	1.3679	.5394	-.1931	.6169	.8615					
.4201	-1.2110	.3238	1.3799	.5994	-.0740	.6478	.8139					
.4598	-1.2160	.3175	1.3941	.6597	.0589	.6431	.7596					
.4996	-1.2102	.3280	1.3709	.7203	.2003	.7277	.6904					
.5397	-1.0590	.3687	1.2857	.7743	.2704	.7466	.6608					
.5795	-.7573	.4599	1.1161	.8394	.3137	.7617	.6369					
.6197	-.5702	.5035	1.0421	.8996	.2924	.7505	.6547					
.6598	-.3203	.5236	1.0091	.9492	.2286	.7358	.6778					
.6997	-.4418	.5476	.9703	1.0000	.0337	.6782	.7671					
.7493	-.3708	.5651	.9409									
.8353	-.2218	.6086	.8744									
.8791	-.1680	.6211	.8550									
.9212	-.0761	.6482	.8133									
1.0000	.0337	.6782	.7671									

TEST 122 PT 21.2789 PSI CN .9194
 RUN 34 TT 100.4205 K CM -.1096
 POINT 11 RC 13.8100 MILLION CC -.0218
 MACH .7692
 ALPHA 5.9373 DEG

CD1 .06390 CDCOR1 .06326
 CD2 .07129 CDCOR2 .07040
 CD3 .06683 CDCOR3 .06598
 CD4 .05179 CDCOR4 .05120
 CD5 .04371 CDCOR5 .04325

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _s L/PT	MLOC	X/C	CP	P _s L/PT	MLOC	X/C	Y/B/2	CP	P _s L/PT	MLOC
0.0000	.2629	.7486	.6577	0.0000	.2629	.7486	.6577	.0500	-.3375	-1.0633	.3797	1.2639
.0083	-.9851	.4825	1.0773	.0052	1.0318	.9633	.2321	.3957	-.3375	-1.2204	.3397	1.3455
.0097	-.9336	.4456	1.2141	.0098	.9116	.9307	.3226	.5003	-.3375	-.6987	.4801	1.0813
.0203	-1.1120	.3620	1.2992	.0200	.7720	.8918	.4085	.6508	-.3375	-.5697	.5110	1.0296
.0300	-1.1582	.3505	1.3229	.0500	.5603	.8306	.5227	.7003	-.3375	-.4912	.5385	.9849
.0400	-1.2112	.3301	1.3662	.0813	.4116	.7940	.5846					
.0608	-1.2622	.3300	1.3664	.1199	.3279	.7675	.6276					
.0800	-1.2313	.3307	1.3650	.1796	.2023	.7320	.6838					
.1000	-1.2597	.3219	1.3841	.2397	.1093	.7072	.7223					
.1997	-1.2660	.3228	1.3822	.2995	.0220	.6828	.7600					
.2500	-1.2992	.3167	1.3959	.3588	-.0641	.6604	.7944					
.2994	-1.2786	.3133	1.4035	.4193	-.1491	.6315	.8389					
.3402	-1.2975	.3132	1.4038	.4793	-.1940	.6219	.8537					
.3795	-1.2442	.3263	1.3746	.5394	-.1961	.6202	.8563					
.4201	-1.0008	.3943	1.2356	.5994	-.0811	.6523	.8069					
.4598	-.7526	.4682	1.1017	.6597	.0821	.6952	.7410					
.4996	-.6627	.4906	1.0635	.7203	.1859	.7281	.6899					
.5397	-.6244	.4990	1.0495	.7743	.2443	.7431	.6663					
.5795	-.6163	.5061	1.0376	.8394	.2820	.7564	.6453					
.6197	-.5881	.5129	1.0265	.8996	.2704	.7525	.6514					
.6598	-.3508	.5308	.9972	.9492	.1667	.7284	.6894					
.6997	-.4714	.5467	.9716	1.0000	-.0699	.6659	.7861					
.7493	-.4075	.5612	.9486									
.8353	-.3394	.5827	.9147									
.8791	-.1958	.6279	.8444									
.9212	-.2563	.6045	.8807									
1.0000	-.0699	.6658	.7861									

TEST 122
 RUN 41
 MACH .786
 R 30.0×10^6



ORIGINAL PAGE IS
OF POOR QUALITY

TEST RUN POINT	122 41 1	PT TT RC MACH ALPHA	51.6765 109.4466 29.8340 -1.9949	PSI K MILLION DEG	CN CM CC	-0.136 -0.098 .0045	CD1 CD2 CD3 CD4 CD5	.00695 .00688 .01741 .00677 .00659	CDCOR1 CDCOR2 CDCOR3 CDCOR4 CDCOR5	.00688 .00680 .01733 .00671 .00657		
UPPER SURFACE												
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	SPANWISE				
0.0000	1.1428	.9343	.0908	0.0000	1.1428	.9343	.0908	X/C	Y/8/Z	CP	P/L/PT	MLOC
.0003	.7203	.8738	.4444	.0052	-.9324	.4049	1.2167	.0500	-.3375	-.0711	.6904	.7492
.0097	.7352	.8787	.4348	.0098	-.7220	.4613	1.1149	.3957	-.3375	-.3188	.3789	.9218
.0203	.4405	.8077	.5623	.0200	-.4901	.5277	1.0036	.5008	-.3375	-.3906	.3568	.9567
.0300	.2589	.7416	.6695	.0500	-.4225	.5455	.9748	.6048	-.3375	-.4649	.3391	.9848
.0400	.1794	.7167	.7684	.0813	-.4690	.5369	.9886	.7003	-.3375	-.4365	.5465	.9732
.0608	.0630	.6879	.7530	.1199	-.4340	.5458	.9743					
.0800	-.0009	.6690	.7821	.1796	-.4715	.5353	.9912					
.1000	-.0721	.6489	.8131	.2397	-.4972	.5150	1.0242					
.1997	-.1447	.6166	.8629	.2995	-.5418	.4985	1.0517					
.2500	-.2336	.6044	.8819	.3588	-.6072	.4798	1.0831					
.2994	-.2762	.5912	.9024	.4193	-.6641	.4929	1.0611					
.3402	-.2895	.5869	.9092	.4793	-.6201	.4929	1.0611					
.3795	-.3130	.5834	.9146	.5394	-.6037	.5578	.9551					
.4201	-.3333	.5776	.9238	.5994	-.2035	.6143	.8665					
.4598	-.3507	.5630	.9468	.6507	-.0126	.6673	.7847					
.4996	-.3983	.5579	.9550	.7203	.1361	.7094	.7198					
.5397	-.4362	.5455	.9748	.7743	.2218	.7322	.6842					
.5795	-.4663	.5379	.9870	.8394	.2809	.7500	.6562					
.6197	-.4715	.5377	.9873	.8996	.3025	.7569	.6453					
.6598	-.4486	.5420	.9804	.9492	.2748	.7495	.6370					
.6997	-.4364	.5456	.9746	1.0000	.2144	.7314	.6854					
.7493	-.3657	.5678	.9392									
.7953	-.1884	.6166	.8629									
.8791	-.0740	.6488	.8131									
.9212	.0186	.6756	.7720									
1.0000	.2144	.7314	.6854									

TEST RUN POINT	122 41 2	PT TT RC MACH ALPHA	51.6857 109.3812 29.8690 -.9614	PSI K MILLION DEG	CN CM CC	-1.474 -1.042 .0073	CD1 CD2 CD3 CD4 CD5	.00678 .00665 .01683 .00658 .00649	CDCOR1 CDCOR2 CDCOR3 CDCOR4 CDCOR5	.00668 .00651 .01669 .00651 .00643		
UPPER SURFACE												
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	SPANWISE				
0.0000	1.1579	.9990	.0383	0.0000	1.1579	.9990	.0383	X/C	Y/8/Z	CP	P/L/PT	MLOC
.0083	.5938	.8389	.5085	.0052	-.2906	.5885	.9067	.0500	-.3375	-.0721	.6511	.8097
.0037	.5461	.8256	.5318	.0098	-.2335	.6041	.8824	.3957	-.3375	-.4013	.3989	.9533
.0203	.2999	.7526	.6521	.0200	-.1819	.6177	.8613	.5008	-.3375	-.4763	.3584	.9920
.0300	.0718	.6898	.7500	.0500	-.1940	.6148	.8657	.6048	-.3375	-.5352	.3174	1.0203
.0400	-.0059	.6683	.7832	.0813	-.2747	.5910	.9028	.7003	-.3375	-.4686	.5384	.9862
.0608	-.0982	.6412	.8249	.1199	-.2671	.5935	.8988					
.0800	-.1549	.6254	.8493	.1796	-.3265	.5773	.9242					
.1000	-.2169	.6085	.8755	.2397	-.3662	.5699	.9360					
.1997	-.2989	.5857	.9111	.2995	-.4207	.5511	.9658					
.2500	-.3418	.5727	.9315	.3588	-.4926	.5298	1.0001					
.2994	-.3739	.5620	.9485	.4193	-.5382	.5151	1.0241					
.3402	-.3819	.5604	.9509	.4793	-.5161	.5223	1.0124					
.3795	-.3997	.5550	.9596	.5394	-.3776	.5612	.9496					
.4201	-.4188	.5500	.9676	.5994	-.1818	.6174	.8617					
.4598	-.4593	.5419	.9821	.6507	.0041	.6723	.7771					
.4996	-.4746	.5346	.9923	.7203	.1572	.7143	.7122					
.5397	-.5131	.5228	1.0114	.7743	.2416	.7377	.6755					
.5795	-.5422	.5174	1.0203	.8394	.2951	.7546	.6489					
.6197	-.5548	.5094	1.0335	.8996	.3171	.7584	.6429					
.6598	-.5135	.5267	1.0052	.9492	.2806	.7512	.6543					
.6997	-.4654	.5441	.9835	1.0000	.2120	.7302	.6874					
.7493	-.3865	.5626	.9475									
.7953	-.1938	.6161	.8636									
.8791	-.0770	.6490	.8129									
.9212	.0184	.6745	.7735									
1.0000	.2120	.7302	.6874									

TEST RUN POINT	122 41 3	PT TT RC MACH ALPHA	51.6860 109.2541 30.0330 .7851 .0300	PSI K MILLION DEG	CN CM CC	.2944 -1.081 .0061	CD1 CD2 CD3 CD4 CD5	.00679 .00687 .01773 .00667 .00650	CDCOR1 CDCOR2 CDCOR3 CDCOR4 CDCOR5	.00673 .00669 .01766 .00661 .00647		
UPPER SURFACE												
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	SPANWISE				
0.0000	1.1477	.9950	.0851	0.0000	1.1477	.9950	.0851	X/C	Y/8/Z	CP	P/L/PT	MLOC
.0083	.3731	.7722	.6207	.0052	.1478	.7092	.7201	.0500	-.3375	-.2189	.6046	.8815
.0397	.3275	.7606	.6394	.0098	.0917	.6931	.7450	.3957	-.3375	-.4930	.5261	1.0061
.0203	.0486	.6800	.7652	.0200	.0654	.6848	.7579	.5008	-.3375	-.5612	.5054	1.0400
.0300	-.1237	.6305	.8414	.0500	-.0109	.6635	.7907	.6048	-.3375	-.6255	.4866	1.0715
.0400	-.1968	.6102	.8728	.0813	-.1192	.6331	.8375	.7003	-.3375	-.4971	.5233	1.0107
.0608	-.2749	.5885	.9066	.1199	-.1320	.6290	.8438					
.0800	-.3163	.5762	.9260	.1796	-.2123	.6080	.8763					
.1000	-.3762	.5613	.9496	.2397	-.2657	.5909	.9029					
.1997	-.4261	.5480	.9707	.2995	-.3237	.5772	.9245					
.2500	-.4540	.5379	.9870	.3588	-.3953	.5547	.9601					
.2994	-.4790	.5303	.9993	.4193	-.4427	.5407	.9825					
.3402	-.4909	.5287	1.0019	.4793	-.4470	.5384	.9862					
.3795	-.4898	.5261	1.0060	.5394	-.3397	.5691	.9371					
.4201	-.4991	.5253	1.0073	.5994	-.1558	.6234	.8525					
.4598	-.5423	.5128	1.0279	.6507	.0258	.6730	.7728					
.4996	-.5530	.5080	1.0358	.7203	.1754	.7187	.7085					
.5397	-.5839	.5020	1.0456	.7743	.2582	.7421	.6686					
.5795	-.6219	.4897	1.0663	.8394	.3099	.7560	.6466					
.6197	-.6325	.4871	1.0707	.8996	.3268	.7611	.6386					
.6598	-.5910	.4999	1.0492	.9492	.2903	.7496	.6568					
.6997	-.4990	.5242	1.0092	1.0000	.2064	.7269	.6926					
.7493	-.3590	.5563	.9575									
.7953	-.1912	.6112	.8712									
.8791	-.0739	.6448	.8155									
.9212	.0223	.6726	.7766									
1.0000	.2064	.7269	.6926									

TEST 122	PT	51.6898	PSI	CN	.4352	CD1	.00693	CDCOR1	.00680
RUN 41	TT	109.5888	K	CM	-.1091	CD2	.00695	CDCOR2	.00671
POINT 4	RC	29.7920	MILLION	CC	.0015	CD3	.01676	CDCOR3	.01654
	MACH	.7807				CD4	.00684	CDCOR4	.00674
	ALPHA	1.0100	DEG			CD5	.00660	CDCOR5	.00654

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC	X/C	Y/B/2	CP	P _L /PT	MLOC
0.0000	1.0668	.9730	.1987	0.0000	1.0668	.9730	.1987	.0500	-.3375	-.3922	.5584	.9540
.0083	.1312	.7671	.7232	.0052	.4489	.7970	.5802	.3957	-.3375	-.5714	.5097	1.0330
.0097	.0848	.6934	.7446	.0098	.3481	.7678	.6277	.5008	-.3375	-.6244	.4936	1.0598
.0203	-.2233	.6050	.8809	.0230	.2716	.7460	.6625	.6048	-.3375	-.6561	.4857	1.0731
.0300	-.3508	.5686	.9379	.0500	.1428	.7092	.7201	.7003	-.3375	-.4931	.5321	.9963
.0400	-.4221	.5482	.9755	.0813	.0176	.6759	.7715					
.0608	-.4930	.5340	.9932	.1199	-.0134	.6646	.7899					
.0800	-.5046	.5245	1.0086	.1796	-.1052	.6387	.8288					
.1000	-.5168	.5100	1.0325	.2397	-.1658	.6225	.8538					
.1197	-.5289	.5069	1.0376	.2995	-.2334	.6048	.8812					
.1500	-.5403	.5070	1.0374	.3588	-.3048	.5826	.9160					
.1694	-.5493	.5028	1.0445	.4193	-.3484	.5727	.9315					
.1802	-.5520	.5024	1.0450	.4793	-.3668	.5663	.9416					
.1945	-.5595	.5063	1.0386	.5394	-.2916	.5879	.9076					
.2001	-.5709	.5087	1.0346	.5994	-.1280	.6343	.8355					
.2098	-.6098	.4983	1.0519	.6507	.0447	.6837	.7595					
.2196	-.6182	.4969	1.0542	.7203	.1903	.7256	.6946					
.2297	-.6490	.4864	1.0718	.7743	.2700	.7474	.6603					
.2397	-.6824	.4767	1.0883	.8394	.3196	.7610	.6386					
.2497	-.6679	.4825	1.0784	.8996	.3334	.7659	.6308					
.2598	-.5925	.5123	1.0453	.9492	.2926	.7534	.6507					
.2697	-.4912	.5327	.9954	1.0000	.2001	.7274	.6917					
.2793	-.3896	.5605	.9507									
.2853	-.1946	.6161	.8637									
.2791	-.0760	.6493	.8124									
.9212	.0212	.6771	.7697									
1.0000	.2001	.7274	.6917									

TEST 122	PT	55.6730	PSI	CN	.5953	CD1	.00745	CDCOR1	.00730
RUN 41	TT	114.6365	K	CM	-.1106	CD2	.00749	CDCOR2	.00722
POINT 5	RC	29.9860	MILLION	CC	-.0069	CD3	.01955	CDCOR3	.01933
	MACH	.7815				CD4	.00728	CDCOR4	.00717
	ALPHA	2.0000	DEG			CD5	.00711	CDCOR5	.00706

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC	X/C	Y/B/2	CP	P _L /PT	MLOC
0.0000	.9104	.9284	.3281	0.0000	.9104	.9284	.3281	.0500	-.3375	-.5644	.5113	1.0301
.0083	-.1178	.6359	.8327	.0052	.6790	.8621	.4662	.3957	-.3375	-.7536	.4574	1.1214
.0097	-.1812	.6169	.8621	.0098	.5481	.8257	.5314	.5008	-.3375	-.7404	.4561	1.1237
.0203	-.4703	.5363	.9892	.0200	.4371	.7916	.5888	.6048	-.3375	-.6349	.4878	1.0692
.0300	-.5489	.5090	1.0338	.0500	.2786	.7498	.6561	.7003	-.3375	-.4870	.5319	.9964
.0400	-.6527	.4958	1.0726	.0813	.1356	.7068	.7234					
.0608	-.6936	.4702	1.0992	.1199	.0900	.6961	.7400					
.0800	-.7230	.4655	1.1073	.1796	-.0155	.6654	.7873					
.1000	-.7535	.4557	1.1245	.2397	-.0846	.6441	.8202					
.1197	-.7935	.4433	1.1462	.2995	-.1552	.6250	.8496					
.1500	-.8081	.4369	1.1576	.3588	-.2285	.6025	.8845					
.1694	-.8001	.4383	1.1552	.4193	-.2785	.5875	.9079					
.1802	-.7943	.4408	1.1507	.4793	-.3050	.5806	.9188					
.1945	-.7798	.4483	1.1373	.5394	-.2490	.5992	.8897					
.2001	-.7628	.4505	1.1336	.5994	-.0994	.6398	.8268					
.2098	-.7554	.4557	1.1244	.6507	.0645	.6885	.7518					
.2196	-.7790	.4423	1.1480	.7203	.2095	.7260	.6936					
.2297	-.7235	.4628	1.1120	.7743	.2855	.7502	.6556					
.2397	-.7026	.4604	1.1058	.8394	.3325	.7624	.6361					
.2497	-.6539	.4875	1.0697	.8996	.3414	.7687	.6260					
.2598	-.5688	.5083	1.0350	.9492	.2937	.7534	.6506					
.2697	-.4650	.5347	.9918	1.0000	.1957	.7248	.6954					
.2793	-.3866	.5610	.9497									
.2853	-.1910	.6167	.8625									
.2791	-.0715	.6491	.8124									
.9212	.0240	.6775	.7688									
1.0000	.1957	.7248	.6954									

TEST 122	PT	55.6453	PSI	CN	.7430	CD1	.01256	CDCOR1	.01196
RUN 41	TT	114.2169	K	CM	-.1183	CD2	.01309	CDCOR2	.01255
POINT 6	RC	30.0970	MILLION	CC	-.0132	CD3	.02718	CDCOR3	.02661
	MACH	.7826				CD4	.01335	CDCOR4	.01257
	ALPHA	2.9800	DEG			CD5	.01390	CDCOR5	.01329

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC	X/C	Y/B/2	CP	P _L /PT	MLOC
0.0000	.7323	.F786	.4349	0.0000	.7323	.8786	.4349	.0500	-.3375	-.6851	.4761	1.0892
.0083	-.2699	.5946	.8968	.0052	.8251	.9040	.3332	.3957	-.3375	-.9420	.3997	1.2263
.0097	-.3995	.5553	.9588	.0098	.6911	.8663	.4584	.5008	-.3375	-.9943	.3845	1.2556
.0203	-.6260	.4921	1.0620	.0200	.5547	.8257	.5314	.6048	-.3375	-.9875	.3837	1.2572
.0300	-.7138	.4626	1.1123	.0500	.3836	.7782	.6108	.7003	-.3375	-.3707	.5630	.9466
.0400	-.7912	.4430	1.1458	.0813	.2321	.7348	.6799					
.0608	-.8528	.4255	1.1783	.1199	.1712	.7162	.7089					
.0800	-.8595	.4214	1.1859	.1796	.0611	.6844	.7575					
.1000	-.8988	.4104	1.2062	.2397	-.0142	.6645	.7888					
.1197	-.9245	.4030	1.2200	.2995	-.0873	.6424	.8228					
.1500	-.9435	.3971	1.2312	.3588	-.1718	.6179	.8606					
.1694	-.9756	.3889	1.2471	.4193	-.2248	.6034	.8831					
.1802	-.9680	.3886	1.2476	.4793	-.2604	.5915	.9016					
.1945	-.9818	.3920	1.2410	.5394	-.2091	.6112	.8709					
.2001	-.9639	.3956	1.2379	.5994	-.0798	.6457	.8177					
.2098	-.9899	.3847	1.2551	.6507	.0790	.6902	.7492					
.2196	-.9711	.3921	1.2408	.7203	.2186	.7311	.6857					
.2297	-.9640	.3863	1.2521	.7743	.2936	.7487	.6580					
.2397	-.9754	.3917	1.2417	.8394	.3370	.7652	.6317					
.2497	-.9587	.3958	1.2339	.8996	.3482	.7681	.6271					
.2598	-.5115	.5216	1.0131	.9492	.3002	.7535	.6504					
.2697	-.3722	.5624	.9467	1.0000	.1839	.7224	.6992					
.2793	-.3049	.5825	.9156									
.2853	-.1596	.6246	.8563									
.2791	-.0530	.6524	.8074									
.9212	.0292	.6742	.7678									
1.0000	.1839	.7224	.6992									

TEST	122	PT	55.6457	PSI		CN	.7913			CD1	.01693	CDCOR1	.01596
RUN	41	TT	114.8764	K		CM	-1.205			CD2	.01714	CDCOR2	.01612
POINT	7	RC	29.8800	MILLION		CC	-.0142			CD3	.03563	CDCOR3	.03455
		MACH	.7832							CD4	.01899	CDCOR4	.01810
		ALPHA	3.4700	DEG						CD5	.01990	CDCOR5	.01840

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _s L/PT	MLOC	X/C	CP	P _s L/PT	MLOC	X/C	Y/B/2	CP	P _s L/PT	MLOC
0.0000	.7069	.8671	.4569	0.0000	.7069	.8671	.4569	.0500	-.3375	-.7150	.4617	1.1138
.0083	-.2487	.5793	.9208	.0052	.8637	.9136	.3624	.3957	-.3375	-.9997	.3818	1.2608
.0097	-.4534	.5553	.9009	.0098	.7327	.9769	.4381	.5008	-.3375	-1.0266	.3713	1.2816
.0203	-.6513	.4728	1.0967	.0200	.5944	.8355	.5143	.6048	-.3375	-1.0757	.3643	1.2957
.0300	-.7559	.4405	1.1405	.0500	.4177	.7885	.5940	.7003	-.3375	-.3856	.5608	.9499
.0400	-.8566	.4262	1.1771	.0813	.2670	.7441	.6652					
.0608	-.8991	.4111	1.2050	.1199	.2024	.7261	.6934					
.0800	-.9314	.4028	1.2205	.1796	.0873	.6929	.7450					
.1000	-.9631	.3930	1.2392	.2397	.0032	.6693	.7813					
.1997	-.9838	.3872	1.2503	.2995	-.0668	.6490	.8126					
.2500	-.9989	.3811	1.2621	.3588	-.1607	.6210	.8557					
.2994	-1.0209	.3736	1.2776	.4193	-.2239	.6021	.8851					
.3402	-1.0292	.3723	1.2796	.4793	-.2528	.5943	.8972					
.3795	-1.0317	.3691	1.2860	.5394	-.2224	.6015	.8860					
.4201	-1.0250	.3731	1.2780	.5994	-.0767	.6447	.8192					
.4598	-1.0120	.3653	1.2937	.6507	-.0692	.6800	.7549					
.4996	-1.0518	.3682	1.2879	.7203	.2172	.7302	.6870					
.5397	-1.0476	.3670	1.2903	.7743	.2877	.7492	.6571					
.5795	-1.0527	.3620	1.3004	.8394	.3319	.7603	.6395					
.6197	-1.0064	.3853	1.2540	.8996	.3416	.7676	.6278					
.6598	-.6185	.4957	1.0558	.9492	.2812	.7507	.6547					
.6997	-.4060	.5515	.9648	1.0000	.1658	.7164	.7085					
.7493	-.2496	.5872	.9484									
.7993	-.1499	.6281	.9449									
.8791	-.0567	.6542	.8845									
.9212	.0232	.6745	.7733									
1.0000	.1658	.7164	.7085									

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TEST	122	PT	55.6584	PSI		CN	.8588			CD1	.02303	CDCOR1	.02190
RUN	41	TT	114.3433	K		CM	-1.246			CD2	.02344	CDCOR2	.02220
POINT	8	RC	30.0450	MILLION		CC	-.0170			CD3	.05321	CDCOR3	.05200
		MACH	.7809							CD4	.02622	CDCOR4	.02585
		ALPHA	3.9600	DEG						CD5	.02581	CDCOR5	.02594

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _s L/PT	MLOC	X/C	CP	P _s L/PT	MLOC	X/C	Y/B/2	CP	P _s L/PT	MLOC
0.0000	.6017	.8392	.5077	0.0000	.6017	.8392	.5077	.0500	-.3375	-.7672	.4486	1.1368
.0083	-.2497	.5613	.9491	.0052	.9167	.9298	.3248	.3957	-.3375	-1.0624	.3687	1.2869
.0097	-.4599	.5697	1.0326	.0098	.7857	.8945	.4030	.5008	-.3375	-1.0945	.3578	1.3089
.0203	-.8206	.4415	1.1495	.0200	.6402	.8525	.4840	.6048	-.3375	-1.1788	.3368	1.3528
.0300	-.8729	.4237	1.1816	.0500	.4419	.7918	.5885	.7003	-.3375	-.4163	.5523	.9636
.0400	-.8771	.4121	1.2030	.0813	.3030	.7561	.6462					
.0608	-.9682	.3950	1.2354	.1199	.2290	.7336	.6818					
.0800	-.9830	.3878	1.2492	.1796	.1178	.7070	.7232					
.1000	-1.0585	.3760	1.2722	.2397	.0323	.6786	.7671					
.1997	-1.0730	.3724	1.2795	.2995	-.0427	.6621	.7925					
.2500	-1.0558	.3674	1.2894	.3588	-.1411	.6282	.8446					
.2994	-1.1197	.3594	1.3056	.4193	-.1652	.6278	.8453					
.3402	-1.0999	.3593	1.3058	.4793	-.2337	.6048	.8809					
.3795	-1.0948	.3544	1.3159	.5394	-.2074	.6081	.8758					
.4201	-1.1091	.3542	1.3164	.5994	-.0810	.6465	.8184					
.4598	-1.0963	.3525	1.3198	.6507	.0737	.6877	.7531					
.4996	-1.0940	.3525	1.3198	.7203	.2135	.7274	.6914					
.5397	-1.1118	.3445	1.3366	.7743	.2831	.7460	.6622					
.5795	-1.1454	.3462	1.3331	.8394	.3310	.7647	.6325					
.6197	-1.1118	.3485	1.3283	.8996	.3330	.7621	.6366					
.6598	-.5738	.5039	1.0423	.9492	.2771	.7469	.6608					
.6997	-.4188	.5461	.9735	1.0000	.1363	.7095	.7193					
.7493	-.3253	.5730	.9307									
.7993	-.1488	.6280	.8451									
.8791	-.0606	.6558	.8022									
.9212	.0262	.6807	.7638									
1.0000	.1363	.7095	.7193									

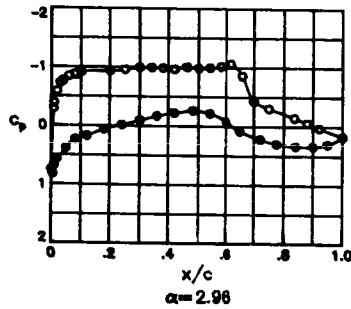
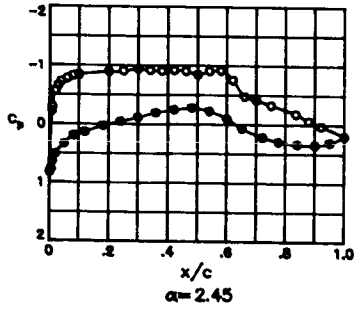
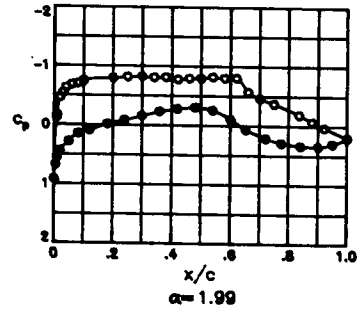
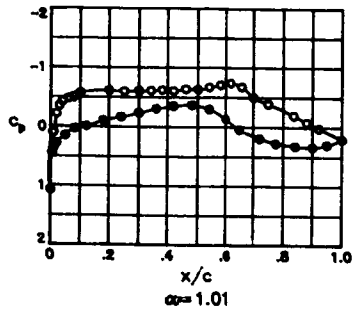
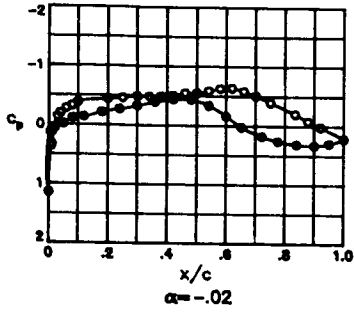
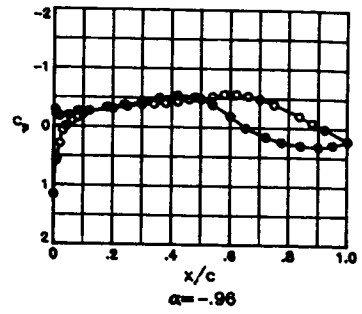
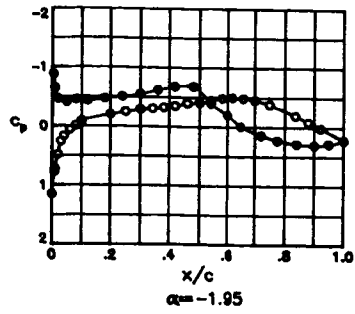
TEST	122	PT	55.6610	PSI		CN	.8921			CD1	.03171	CDCOR1	.03063
RUN	41	TT	114.4782	K		CM	-1.295			CD2	.03308	CDCOR2	.03191
POINT	9	RC	30.0340	MILLION		CC	-.0154			CD3	.06880	CDCOR3	.06770
		MACH	.7832							CD4	.03975	CDCOR4	.03935
		ALPHA	4.4500	DEG						CD5	.03516	CDCOR5	.03488

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _s L/PT	MLOC	X/C	CP	P _s L/PT	MLOC	X/C	Y/B/2	CP	P _s L/PT	MLOC
0.0000	.5233	.8179	.5448	0.0000	.5233	.8179	.5448	.0500	-.3375	-.8511	.4304	1.1695
.0083	-.4248	.5478	.9708	.0052	.9710	.9469	.2806	.3957	-.3375	-1.0870	.3537	1.3174
.0097	-.6910	.4788	1.0845	.0098	.8276	.9052	.3805	.5008	-.3375	-1.0970	.3430	1.3397
.0203	-.8792	.4208	1.1869	.0200	.6864	.8662	.4586	.6048	-.3375	-.8454	.4273	1.1752
.0300	-.9680	.3989	1.2279	.0500	.4886	.8073	.5627	.7003	-.3375	-.4522	.5441	.9767
.0400	-.9715	.3963	1.2443	.0813	.3370	.7685	.6264					
.0608	-1.0533	.3769	1.2704	.1199	.2574	.7429	.6670					
.0800	-1.0363	.3752	1.2738	.1796	.1344	.7061	.7246					
.1000	-1.0561	.3660	1.2923	.2397	.0556	.6854	.7565					
.1997	-1.0946	.3587	1.3670	.2995	-.0293	.6615	.7934					
.2500	-1.1317	.3541	1.3166	.3588	-.1033	.6440	.8203					
.2994	-1.1255	.3463	1.3327	.4193	-.1792	.6185	.8628					
.3402	-1.1187	.3469	1.3315	.4793	-.2325	.6005	.8876					
.3795	-1.1574	.3424	1.3409	.5394	-.1980	.6145	.8658					
.4201	-1.1677	.3389	1.3485	.5994	-.0770	.6485	.8134					
.4598	-1.1225	.3348	1.3572	.6507	.0623	.6789	.7666					
.4996	-1.1519	.3397	1.3467	.7203	.2096	.7278	.6908					
.5397	-1.1611	.3310	1.3655	.7743	.2794	.7453	.6634					
.5795	-1.0674	.3523	1.3263	.8394	.3086	.7511	.6541					
.6197	-.7400	.4543	1.1269	.8996	.3140	.7564	.6458					
.6598	-.5125	.5185	1.0183	.9492	.2450	.7360	.6780					
.6997	-.4435	.5306	.9844	1.0000	-.0114	.6674	.7843					
.7493	-.3579	.5640	.9440									
.7993	-.1941	.6162	.8633									
.8791	-.1247	.6328	.8376									
.9212	-.1459	.6214	.8552									
1.0000	-.0114	.6674	.7843									

TEST	122	PT	55.6620	PSI	CN	.R976	CD1	.04141	CDCDR1	.04032
RUN	41	TT	114.7245	K	CM	-1263	CD2	.04431	CDCDR2	.04301
PDINT	1C	RC	29.9000	MILLION	CC	-0156	CD3	.09154	CDCDR3	.09020
		MACH	1.7862	DEG			CD4	.04433	CDCDR4	.04324
		ALPHA	4.9500				CD5	.03908	CDCDR5	.03817

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/Z	CP	P/L/PT	MLOC
0.0000	.4536	.7989	.5767	0.0000	.4536	.7989	.5767	.0500	-.3375	-.8847	.4182	1.1917
.0083	-.4844	.5325	.9454	.0052	.9886	.9504	.2711	.3957	-.3375	-1.1669	.3438	1.3380
.0097	-.7404	.4576	1.1212	.0098	.8569	.9137	.3620	.5008	-.3375	-1.1812	.3345	1.3578
.0203	-.9337	.4060	1.2145	.0200	.7090	.8709	.4497	.6048	-.3375	-.6980	.4729	1.0945
.0300	-.9739	.3916	1.2418	.0500	.5126	.8155	.5489	.7003	-.3375	-.4587	.5382	.9861
.0400	-1.0218	.3794	1.2657	.0813	.3497	.7662	.6301					
.0608	-1.0179	.3737	1.2768	.1199	.2747	.7434	.6663					
.0800	-1.0609	.3587	1.3071	.1796	.1503	.7091	.7199					
.1000	-1.0648	.3548	1.3151	.2397	-.0709	.6903	.7491					
.1997	-1.1390	.3491	1.3270	.2995	-.0150	.6672	.7846					
.2500	-1.1444	.3431	1.3394	.3588	-.1054	.6390	.8280					
.2994	-1.1431	.3366	1.3535	.4193	-.1851	.6119	.8699					
.3402	-1.1736	.3359	1.3550	.4793	-.2167	.6080	.8760					
.3795	-1.1945	.3317	1.3639	.5394	-.1965	.6148	.8654					
.4201	-1.1750	.3281	1.3716	.5994	-.0964	.6379	.8297					
.4598	-1.1977	.3262	1.3892	.6507	.0558	.6808	.7636					
.4996	-1.2008	.3282	1.3715	.7203	.2059	.7282	.6902					
.5397	-1.1039	.3529	1.3190	.7743	.2673	.7443	.6649					
.5795	-.6685	.4747	1.0915	.8394	.2982	.7518	.6530					
.6197	-.5591	.5118	1.0293	.8996	.2981	.7550	.6479					
.6598	-.4938	.5239	1.0093	.9492	.2305	.7319	.6844					
.6997	-.4590	.5366	.9887	1.0000	.0496	.6874	.7535					
.7493	-.3889	.5662	.9415									
.8353	-.2663	.5945	.8970									
.8791	-.1722	.6234	.8521									
.9212	-.1643	.6154	.8644									
1.0000	.0496	.6874	.7535									

TEST 122
 RUN 53
 MACH .788
 R 45.0×10^6



TEST	122	PT	72.0643	PSI	CN	-0.0023	CD1	.00648	CDCOR1	.00648
RUN	53	TT	104.2614	K	CM	-1.025	CD2	.00643	CDCOR2	.00642
POINT	1	RC	44.8310	MILLION	CC	.0053	CD3	.00640	CDCOR3	.00638
		MACH	.7838				CD4	.00628	CDCOR4	.00629
		ALPHA	-1.9472	DEG			CD5	.00609	CDCOR5	.00611

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/2	CP	P/L/PT	MLOC
0.0000	1.1523	.9965	.0713	0.0000	1.1523	.9965	.0713	.0500	-.3375	.0579	.6852	.7588
.0083	.7752	.8887	.4160	.0052	-.8798	.4139	1.2021	.3957	-.3375	-.3292	.5751	.9295
.0097	.7297	.8752	.4427	.0098	-.6459	.4618	1.0817	.5008	-.3375	-.4135	.5519	.9664
.0203	.4323	.8047	.5686	.0200	-.4680	.5314	.9995	.6048	-.3375	-.4878	.5306	1.0008
.0300	.2581	.7397	.6739	.0500	-.4085	.5314	.9671	.7003	-.3375	-.4487	.5422	.9820
.0400	.1695	.6769	.7105	.0813	-.4490	.5397	.9860					
.0608	.0590	.6847	.7596	.1199	-.4293	.5444	.9785					
.0800	-.0262	.6653	.7894	.1796	-.4680	.5341	.9950					
.1000	-.0402	.6448	.8211	.2397	-.5009	.5261	1.0081					
.1997	-.1981	.6118	.8722	.2995	-.5472	.5123	1.0307					
.2500	-.2455	.5970	.8951	.3588	-.6185	.4904	1.0671					
.2994	-.2865	.5859	.9126	.4193	-.6779	.4742	1.0947					
.3402	-.3040	.5820	.9186	.4793	-.6816	.4745	1.0942					
.3795	-.3229	.5769	.9268	.5394	-.3911	.5574	.9575					
.4201	-.3474	.5696	.9382	.5994	-.1957	.6128	.8706					
.4598	-.3951	.5550	.9614	.6507	-.0046	.6664	.7877					
.4996	-.4077	.5529	.9649	.7203	-.1465	.7106	.7194					
.5397	-.4476	.5432	.9803	.7743	-.2306	.7357	.6802					
.5795	-.4813	.5331	.9967	.8394	.2884	.7517	.6549					
.6197	-.4928	.5290	1.0033	.8996	.3108	.7576	.6455					
.6598	-.4810	.5315	.9992	.9492	.2861	.7500	.6575					
.6997	-.44501	.5412	.9835	1.0000	.2247	.7325	.6852					
.7493	-.3754	.5631	.9486									
.7933	-.1912	.6147	.8677									
.8791	-.0763	.6481	.8160									
.9212	.0212	.6746	.7751									
1.0000	.2247	.7325	.6852									

TEST	122	PT	72.0643	PSI	CN	.1556	CD1	.00621	CDCOR1	.00621
RUN	53	TT	104.5559	K	CM	-1.069	CD2	.00628	CDCOR2	.00626
POINT	2	RC	44.5660	MILLION	CC	.0074	CD3	.00620	CDCOR3	.00618
		MACH	.7813				CD4	.00620	CDCOR4	.00619
		ALPHA	-.9600	DEG			CD5	.00601	CDCOR5	.00603

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/2	CP	P/L/PT	MLOC
0.0000	1.1617	.9997	.0216	0.0000	1.1617	.9997	.0216	.0500	-.3375	-.0595	.6531	.8082
.0083	.5856	.8360	.5148	.0052	-.2845	.5883	.9088	.3957	-.3375	-.4061	.5550	.9613
.0097	.5458	.8244	.5350	.0098	-.2106	.6088	.8768	.5008	-.3375	-.4835	.5345	.9944
.0203	.2418	.7518	.6546	.0200	-.1786	.6199	.8596	.6048	-.3375	-.5477	.5168	1.0232
.0300	.0741	.6915	.7489	.0500	-.1921	.6145	.8679	.7003	-.3375	-.4824	.5319	.9986
.0400	-.0625	.6884	.7886	.0813	-.2633	.5936	.9004					
.0608	-.0972	.6410	.8269	.1199	-.2672	.5926	.9020					
.0800	-.1521	.6254	.8510	.1796	-.3291	.5747	.9301					
.1000	-.2189	.6061	.8810	.2397	-.3713	.5666	.9430					
.1997	-.3696	.5824	.9180	.2995	-.4268	.5492	.9707					
.2500	-.3473	.5734	.9322	.3588	-.4936	.5320	.9984					
.2994	-.3795	.5619	.9504	.4193	-.5372	.5171	1.0227					
.3402	-.3893	.5637	.9475	.4793	-.5034	.5299	1.0018					
.3795	-.4031	.5563	.9594	.5394	-.3769	.5637	.9475					
.4201	-.4255	.5492	.9707	.5994	-.1824	.6182	.8622					
.4598	-.4716	.5268	.9906	.6507	.0085	.6729	.7776					
.4996	-.4754	.5359	.9921	.7203	.1624	.7167	.7099					
.5397	-.5194	.5217	1.0152	.7743	.2482	.7399	.6735					
.5795	-.5581	.5109	1.0330	.8394	.3013	.7551	.6494					
.6197	-.5604	.5102	1.0341	.8996	.3223	.7610	.6399					
.6598	-.5304	.5205	1.0171	.9492	.2905	.7531	.6526					
.6997	-.4801	.5356	.9926	1.0000	.2188	.7320	.6859					
.7493	-.3681	.5610	.9518									
.7933	-.1976	.6143	.8681									
.8791	-.0776	.6482	.8158									
.9212	.0212	.6752	.7742									
1.0000	.2188	.7320	.6859									

TEST	122	PT	72.0608	PSI	CN	.3004	CD1	.00629	CDCOR1	.00625
RUN	53	TT	104.4996	K	CM	-1.099	CD2	.00618	CDCOR2	.00614
POINT	3	RC	44.5500	MILLION	CC	.0063	CD3	.00618	CDCOR3	.00610
		MACH	.7797				CD4	.00614	CDCOR4	.00612
		ALPHA	-.6198	DEG			CD5	.00600	CDCOR5	.00600

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/2	CP	P/L/PT	MLOC
0.0000	1.1482	.9960	.0757	0.0000	1.1482	.9960	.0757	.0500	-.3375	-.1997	.6165	.8648
.0083	.3693	.7752	.6171	.0052	-.1442	.7099	.7205	.3957	-.3375	-.4931	.5327	.9973
.0097	.3310	.7631	.6367	.0098	-.1041	.6992	.7370	.5008	-.3375	-.5533	.5162	1.0241
.0203	.0488	.6835	.7613	.0200	-.0680	.6892	.7526	.6048	-.3375	-.6056	.5013	1.0489
.0300	-.1281	.6335	.8385	.0500	-.0101	.6679	.7855	.7003	-.3375	-.4973	.5297	1.0021
.0400	-.1965	.6150	.8671	.0813	-.1091	.6386	.8305					
.0608	-.2765	.5911	.9044	.1199	-.1319	.6322	.8405					
.0800	-.3174	.5795	.9225	.1796	-.2120	.6095	.8756					
.1000	-.3791	.5621	.9501	.2397	-.2650	.5965	.8960					
.1997	-.4333	.5493	.9706	.2995	-.3246	.5800	.9218					
.2500	-.4563	.5423	.9817	.3588	-.3914	.5607	.9523					
.2994	-.4842	.5345	.9943	.4193	-.4374	.5469	.9743					
.3402	-.4816	.5345	.9943	.4793	-.4363	.5474	.9736					
.3795	-.4478	.5340	.9952	.5394	-.3371	.5766	.9272					
.4201	-.5038	.5304	1.0010	.5994	-.1598	.6274	.8479					
.4598	-.5452	.5172	1.0226	.6507	.0267	.6790	.7684					
.4996	-.5440	.5185	1.0205	.7203	.1793	.7234	.6994					
.5397	-.5448	.5067	1.0398	.7743	.2620	.7459	.6640					
.5795	-.6238	.4941	1.0609	.8394	.3142	.7598	.6410					
.6197	-.6346	.4834	1.0621	.8996	.3308	.7658	.6323					
.6598	-.5805	.5067	1.0396	.9492	.2958	.7549	.6497					
.6997	-.5053	.5264	1.0275	1.0000	.2164	.7325	.6851					
.7493	-.3923	.5665	.9527									
.7933	-.1984	.6168	.8644									
.8791	-.0766	.6487	.8151									
.9212	.0195	.6773	.7709									
1.0000	.2164	.7325	.6851									

TEST	122	PT	72.0603	PSI	CM	.4545	CD1	.00661	CDCOR1	.00653
RUN	53	TT	104.5386	K	CM	-.1132	CD2	.00660	CDCOR2	.00652
POINT	4	RC	44.6370	MILLION	CC	.0013	CD3	.00664	CDCOR3	.00646
		MACH	.7836				CD4	.00648	CDCOR4	.00641
		ALPHA	1.0100	DEG			CD5	.00614	CDCOR5	.00612

UPPER SURFACE					LOWER SURFACE					SPANWISE				
X/C	CP	P/L/PT	MLOC		X/C	CP	P/L/PT	MLOC		X/C	Y/R/2	CP	P/L/PT	MLOC
0.0000	1.0725	.9741	.1949		0.0000	1.0725	.9741	.1949		.0500	-.3375	-.3609	.5677	.9412
.0083	-.1076	.6993	.7370		.0052	.4525	.7970	.5814		.3957	-.3375	-.6051	.4985	1.0534
.0097	-.0955	.6951	.7434		.0098	.3614	.7710	.6240		.5008	-.3375	-.6389	.4982	1.0700
.0203	-.2101	.6682	.8809		.0200	-.2787	.7469	.6624		.6048	-.3375	-.7133	.4652	1.1100
.0300	-.3479	.5679	.9409		.0500	-.1501	.7112	.7712		.7003	-.3375	-.4936	.5275	1.0058
.0400	-.4421	.5484	.9719		.0813	.0317	.6771	.7892						
.0608	-.4768	.5321	.9982		.1199	-.0095	.6661	.7892						
.0800	-.5094	.5238	1.0117		.1796	-.1049	.6393	.8296						
.1000	-.5648	.5084	1.0370		.2397	-.1664	.6212	.8575						
.1997	-.6094	.4934	1.0620		.2995	-.2334	.6008	.8892						
.2500	-.5966	.4988	1.0531		.3588	-.3054	.5817	.9191						
.2994	-.6036	.4977	1.0548		.4193	-.3513	.5695	.9384						
.3402	-.6096	.4954	1.0586		.4793	-.3675	.5644	.9465						
.3795	-.6103	.4928	1.0631		.5394	-.2943	.5593	.9135						
.4201	-.6152	.4959	1.0578		.5994	-.1252	.6327	.8397						
.4598	-.6354	.4874	1.0722		.6507	.0485	.6823	.7633						
.4996	-.6287	.4906	1.0668		.7233	.1988	.7258	.6956						
.5397	-.6651	.4811	1.0828		.7743	.2784	.7490	.6592						
.5795	-.7045	.4703	1.1014		.8344	.3266	.7628	.6371						
.6197	-.7377	.4586	1.1216		.8996	.3421	.7661	.6318						
.6598	-.6726	.4774	1.0892		.9492	.3022	.7549	.6498						
.6997	-.4429	.5290	1.0032		1.0000	.2084	.7297	.6895						
.7493	-.3821	.5603	.9529											
.8353	-.1891	.6148	.8675											
.8791	-.0722	.6488	.8148											
.9212	.0261	.6760	.7730											
1.0000	.2084	.7297	.6895											

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TEST	122	PT	72.1077	PSI	CM	.6124	CD1	.00755	CDCOR1	.00757
RUN	53	TT	104.5331	K	CM	-.1158	CD2	.00751	CDCOR2	.00754
POINT	8	RC	44.6160	MILLION	CC	-.0064	CD3	.00737	CDCOR3	.00739
		MACH	.7834				CD4	.00739	CDCOR4	.00741
		ALPHA	1.9900	DEG			CD5	.00699	CDCOR5	.00714

UPPER SURFACE					LOWER SURFACE					SPANWISE				
X/C	CP	P/L/PT	MLOC		X/C	CP	P/L/PT	MLOC		X/C	Y/R/2	CP	P/L/PT	MLOC
0.0000	.9236	.9313	.3219		0.0000	.9236	.9313	.3219		.0500	-.3375	-.5107	.5229	1.0132
.0083	-.1224	.6327	.8397		.0052	.6845	.8637	.4644		.3957	-.3375	-.7918	.4439	1.1475
.0097	-.1656	.6218	.8566		.0098	.5558	.8279	.5289		.5008	-.3375	-.7881	.4468	1.1424
.0203	-.4620	.5391	.9869		.0200	.4402	.7927	.5884		.6048	-.3375	-.6935	.4760	1.0915
.0300	-.5391	.5126	1.0301		.0500	.2857	.7493	.6586		.7003	-.3375	-.4452	.5420	.9823
.0400	-.6271	.4888	1.0697		.0813	.1476	.7089	.7220						
.0608	-.6828	.4713	1.0996		.1199	-.0942	.6958	.7454						
.0800	-.6999	.4667	1.1075		.1796	-.0122	.6644	.7907						
.1000	-.7467	.4551	1.1277		.2397	-.0828	.6446	.8214						
.1997	-.7918	.4423	1.1503		.2995	-.1544	.6241	.8531						
.2500	-.8105	.4351	1.1632		.3588	-.2306	.6009	.8890						
.2994	-.8181	.4364	1.1609		.4193	-.2760	.5906	.9051						
.3402	-.8040	.4389	1.1563		.4793	-.3025	.5819	.9187						
.3795	-.8083	.4415	1.1518		.5394	-.2528	.5989	.8922						
.4201	-.7740	.4521	1.1330		.5994	-.1022	.6422	.8250						
.4598	-.7922	.4407	1.1531		.6507	.0707	.6873	.7554						
.4996	-.7932	.4425	1.1490		.7203	.2136	.7294	.6901						
.5397	-.8165	.4367	1.1603		.7743	.2398	.7526	.6534						
.5795	-.7970	.4426	1.1497		.8394	.3389	.7657	.6325						
.6197	-.8006	.4422	1.1506		.8996	.3495	.7689	.6273						
.6598	-.5652	.5091	1.0359		.9492	.3054	.7564	.6473						
.6997	-.4546	.5420	.9823		1.0000	.2032	.7277	.6928						
.7493	-.3671	.5655	.9454											
.8353	-.1858	.6178	.8628											
.8791	-.0648	.6500	.8130											
.9212	.0318	.6777	.7703											
1.0000	.2032	.7277	.6928											

TEST	122	PT	72.1317	PSI	CM	.6857	CD1	.00910	CDCOR1	.00922
RUN	53	TT	104.4820	K	CM	-.1144	CD2	.00922	CDCOR2	.00929
POINT	9	RC	44.5320	MILLION	CC	-.0113	CD3	.00933	CDCOR3	.00940
		MACH	.7802				CD4	.00901	CDCOR4	.00914
		ALPHA	2.4500	DEG			CD5	.00878	CDCOR5	.00896

UPPER SURFACE					LOWER SURFACE					SPANWISE				
X/C	CP	P/L/PT	MLOC		X/C	CP	P/L/PT	MLOC		X/C	Y/R/2	CP	P/L/PT	MLOC
0.0000	.8268	.9044	.3834		0.0000	.8268	.9044	.3834		.0500	-.3375	-.5998	.5041	1.0442
.0083	-.2172	.6676	.8787		.0052	.7661	.8878	.4177		.3957	-.3375	-.8857	.4244	1.1826
.0097	-.2913	.5883	.9088		.0098	.6356	.8508	.4882		.5008	-.3375	-.9360	.4027	1.2230
.0203	-.5413	.5117	1.0316		.0200	.5084	.8154	.5504		.6048	-.3375	-.8141	.4365	1.1608
.0300	-.6549	.4865	1.0736		.0500	.3403	.7669	.6304		.7003	-.3375	-.4109	.5523	.9658
.0400	-.7253	.4648	1.1107		.0813	.1976	.7246	.6975						
.0608	-.7753	.4473	1.1414		.1199	.1390	.7122	.7169						
.0800	-.8208	.4417	1.1514		.1796	.0277	.6778	.7701						
.1000	-.8426	.4307	1.1711		.2397	-.0428	.6595	.7983						
.1997	-.8905	.4221	1.1867		.2995	-.1147	.6407	.8273						
.2500	-.9006	.4165	1.1971		.3588	-.1935	.6166	.8646						
.2994	-.9393	.4104	1.2084		.4193	-.2422	.6062	.8808						
.3402	-.9127	.4138	1.2021		.4793	-.2745	.5942	.8995						
.3795	-.9014	.4209	1.1890		.5394	-.2203	.6122	.8713						
.4201	-.9148	.4151	1.1996		.5994	-.0948	.6462	.8188						
.4598	-.8986	.4162	1.1976		.6507	.0737	.6917	.7487						
.4996	-.8829	.4369	1.1598		.7203	.2201	.7358	.6790						
.5397	-.9005	.4405	1.1970		.7743	.2972	.7554	.6489						
.5795	-.9106	.4150	1.1909		.8394	.3399	.7681	.6245						
.6197	-.7535	.4577	1.1230		.8996	.3517	.7706	.6244						
.6598	-.4747	.5393	.9866		.9492	.3055	.7591	.6430						
.6997	-.4470	.5549	.9616		1.0000	.1995	.7255	.6961						
.7493	-.3228	.5765	.9273											
.8353	-.1768	.6222	.8559											
.8791	-.0583	.6534	.8078											
.9212	.0303	.6803	.7663											
1.0000	.1995	.7255	.6961											

TEST	122	PT	76.2179	PSI	CN	.7697	CD1	.01451	CDCOR1	.61394
RUN	53	TT	108.4508	K	CM	-.1273	CD2	.01589	CDCOR2	.01531
POINT	7	RC	44.6260	MILLION	CC	-.0118	CD3	.01515	CDCOR3	.01496
		MACH	.7844				CD4	.01516	CDCOR4	.01454
		ALPHA	2.9600	DEG			CD5	.01559	CDCOR5	.01504

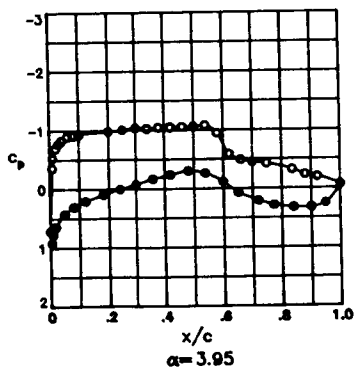
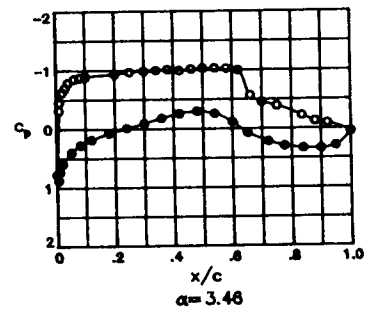
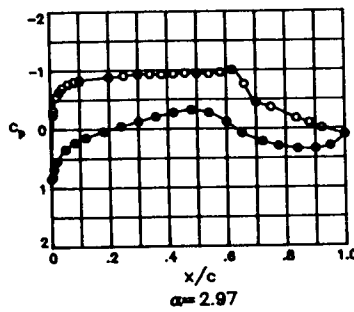
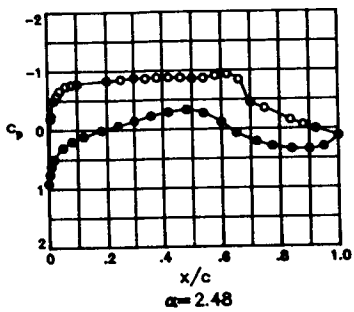
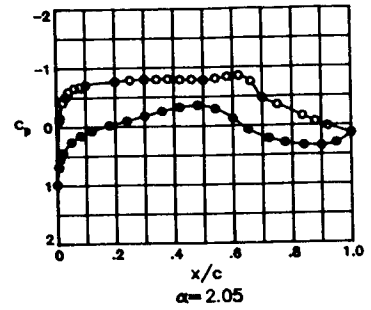
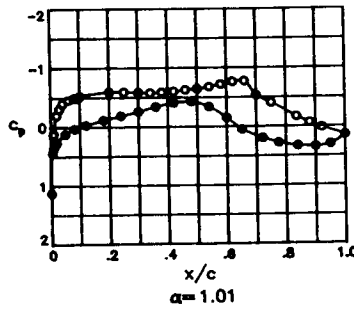
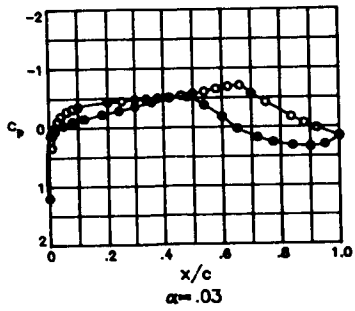
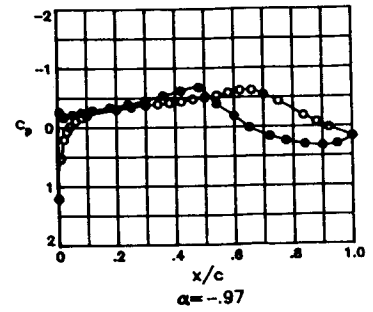
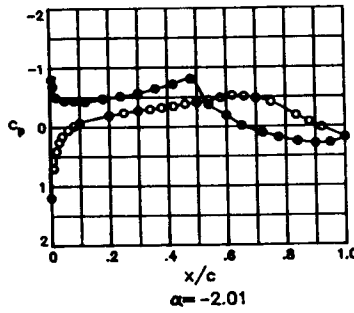
UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _o L/PT	MLOC	X/C	CP	P _o L/PT	MLOC	X/C	Y/S/2	CP	P _o L/PT	MLOC
0.0000	.7566	.8837	.4256	0.0000	.7566	.8837	.4256	.0500	-.3375	-.6128	.4902	1.0671
.0083	-.2897	.5850	.9136	.0052	.8277	.9049	.3820	.3957	-.3375	-.9407	.3969	1.2336
.0097	-.3950	.5603	.9525	.0098	.6825	.8605	.4701	.5008	-.3375	-.9931	.3814	1.2635
.0203	-.5427	.4959	1.0574	.0200	.5643	.8295	.5258	.6048	-.3375	-1.0620	.3665	1.2931
.0300	-.7219	.4633	1.1130	.0500	.3878	.7767	.6142	.7003	-.3375	-.4909	.5285	1.0037
.0400	-.7685	.4448	1.1455	.0813	.2380	.7360	.6793					
.0608	-.8417	.4280	1.1757	.1199	.1782	.7198	.7045					
.0800	-.8733	.4207	1.1890	.1796	.0650	.6865	.7563					
.1000	-.9080	.4088	1.2110	.2397	-.0121	.6652	.7891					
.1497	-.9263	.4018	1.2243	.2995	-.0896	.6412	.8262					
.2500	-.9514	.3963	1.2347	.3588	-.1667	.6203	.8585					
.2994	-.9892	.3875	1.2516	.4193	-.2219	.6059	.8808					
.3402	-.9904	.3879	1.2510	.4793	-.2601	.5955	.8971					
.3795	-.9819	.3900	1.2468	.5394	-.2101	.6095	.8752					
.4201	-.9636	.3921	1.2427	.5994	-.0767	.6455	.8195					
.4598	-.9999	.3811	1.2642	.6507	.0830	.6908	.7497					
.4996	-.9794	.3883	1.2502	.7203	.2252	.7321	.6854					
.5397	-.9891	.3885	1.2497	.7743	.3008	.7551	.6490					
.5795	-1.0109	.3768	1.2726	.8394	.3412	.7641	.6346					
.6197	-1.0647	.3600	1.3065	.8996	.3449	.7645	.6339					
.6598	-.8584	.4239	1.1832	.9492	.3053	.7555	.6484					
.6997	-.4212	.5507	.9679	1.0000	.1787	.7213	.7023					
.7493	-.2983	.5818	.9185									
.8353	-.1414	.6275	.8474									
.8791	-.6528	.6546	.8659									
.9212	.0314	.6757	.7729									
1.0690	.1787	.7213	.7023									

Appendix G

Pressure Data for $M = 0.80$; $R = 4.4 \times 10^6$, 7.7×10^6 , 14.0×10^6 , 30.0×10^6 , and 45.0×10^6 , and 50.0×10^6 ; and Free Transition

The pressure measurements made on the Boeing BAC I airfoil are presented in coefficient form in graphs and tables in this appendix. The data are for a Mach number of 0.80; Reynolds numbers of 4.4×10^6 , 7.7×10^6 , 14.0×10^6 , 30.0×10^6 , 45.0×10^6 , and 50.0×10^6 ; and free transition. The pressure data from the upper surface of the airfoil are plotted as open symbols, and the data from the lower surface are plotted as solid symbols.

TEST 122
 RUN 18
 MACH .807
 R 4.4×10^6



TEST 122	PT	17.7667	PSI	CN	-0.203	CD1	.00832	CDCOR1	.00791
RUN 18	TT	196.9888	K	CM	-0.980	CD2	.00877	CDCOR2	.00835
POINT 1	RC	4.4800	MILLION	CC	-0.069	CD3	.01025	CDCOR3	.00985
	MACH	.8023				CD4	.00694	CDCOR4	.00675
	ALPHA	-2.0144	DEG			CD5	.00651	CDCOR5	.00642

UPPER SURFACE					LOWER SURFACE					SPANWISE				
X/C	CP	P/L/PT	MLOC		X/C	CP	P/L/PT	MLOC		X/C	Y/8/2	CP	P/L/PT	MLOC
0.0000	1.1956	1.0070	0.0000		0.0000	1.1956	1.0070	0.0000		.0500	-.3375	.0454	.6675	.7822
.0083	.6437	.8588	.4712		.0052	-.8063	.4164	1.1924		.3957	-.3375	-.3405	.3541	.9584
.0097	.7604	.8610	.4671		.0098	-.6859	.4528	1.1276		.5008	-.3375	-.4243	.5281	1.0001
.0203	.4656	.7743	.6156		.0200	-.5025	.5056	1.0370		.6048	-.3375	-.5156	.5011	1.0444
.0300	.2533	.7288	.6875		.0500	-.4450	.5231	1.0083		.7003	-.3375	-.4923	.5098	1.0300
.0400	.1573	.7008	.7309		.0813	-.4402	.5239	1.0070						
.0608	.0505	.6688	.7801		.1199	-.4304	.5274	1.0013						
.0800	-.0181	.6490	.8105		.1796	-.4801	.5131	1.0247						
.1000	-.0720	.6334	.8345		.2397	-.5213	.5021	1.0428						
.1997	-.1951	.5969	.8910		.2995	-.5705	.4861	1.0695						
.2500	-.2409	.5838	.9115		.3588	-.6447	.4648	1.1060						
.2994	-.2776	.5737	.9273		.4193	-.7219	.4430	1.1443						
.3402	-.2991	.5683	.9358		.4793	-.8109	.4179	1.1896						
.3795	-.3248	.5602	.9486		.5394	-.9000	.3940	.9745						
.4201	-.3568	.5522	.9613		.5994	-.9929	.3697	.8882						
.4598	-.3883	.5415	.9786		.6597	-.0210	.6496	.8097						
.4996	-.4168	.5334	.9916		.7203	.1016	.6858	.7541						
.5397	-.4549	.5224	1.0094		.7743	.1808	.7092	.7179						
.5795	-.4939	.5117	1.0269		.8394	.2500	.7300	.6856						
.6197	-.5285	.5011	1.0445		.8996	.2863	.7404	.6693						
.6598	-.5143	.5071	1.0346		.9492	.2741	.7379	.6732						
.6997	-.4820	.5136	1.0238		1.0000	.1840	.7080	.7197						
.7493	-.4183	.5308	.9961											
.8353	-.1892	.4998	.8865											
.8791	-.0799	.6324	.8361											
.9212	.0139	.6587	.7956											
1.0000	.1840	.7080	.7197											

ORIGINAL PAGE IS
OF POOR QUALITY

TEST 122	PT	17.7707	PSI	CN	-1.184	CD1	.00731	CDCOR1	.00702
RUN 18	TT	197.0747	K	CM	-1.082	CD2	.00662	CDCOR2	.00631
POINT 2	RC	4.4792	MILLION	CC	.0083	CD3	.00746	CDCOR3	.00717
	MACH	.8034				CD4	.00585	CDCOR4	.00549
	ALPHA	-.9690	DEG			CD5	.00508	CDCOR5	.00501

UPPER SURFACE					LOWER SURFACE					SPANWISE				
X/C	CP	P/L/PT	MLOC		X/C	CP	P/L/PT	MLOC		X/C	Y/8/2	CP	P/L/PT	MLOC
0.0000	1.2084	1.0114	0.0000		0.0000	1.2084	1.0114	0.0000		.0500	-.3375	-.0546	.6371	.8288
.0083	.5343	.8136	.5507		.0052	-.2685	.5782	.9202		.3957	-.3375	-.4254	.5287	.9992
.0097	.5186	.8091	.5584		.0098	-.2134	.5928	.8974		.5008	-.3375	-.5038	.5046	1.0387
.0203	.2630	.7153	.7485		.0200	-.1807	.6027	.8820		.6048	-.3375	-.6005	.4752	1.0881
.0300	.0736	.6775	.7668		.0500	-.2175	.5922	.8983		.7003	-.3375	-.5353	.4960	1.0530
.0400	-.0142	.6520	.8660		.0813	-.2581	.5799	.9176						
.0608	-.1103	.6234	.8501		.1199	-.2930	.5694	.9340						
.0800	-.1656	.6069	.8755		.1796	-.3415	.5538	.9588						
.1000	-.2100	.5926	.8977		.2397	-.3955	.5387	.9829						
.1997	-.3093	.5628	.9444		.2995	-.4536	.5203	1.0129						
.2500	-.3491	.5519	.9618		.3588	-.5379	.4963	1.0525						
.2994	-.3796	.5428	.9764		.4193	-.6172	.4727	1.0924						
.3402	-.3927	.5377	.9886		.4793	-.6749	.4543	1.1243						
.3795	-.4137	.5309	.9987		.5394	-.7342	.4366	.9864						
.4201	-.4359	.5253	1.0048		.5994	-.7962	.4190	.8878						
.4598	-.4726	.5127	1.0253		.6597	.0123	.6563	.7993						
.4996	-.4991	.5057	1.0368		.7203	.1605	.7008	.7309						
.5397	-.5366	.4948	1.0551		.7743	.2397	.7243	.6945						
.5795	-.5821	.4817	1.0771		.8394	.3016	.7428	.6656						
.6197	-.6170	.4714	1.0946		.8996	.3275	.7505	.6335						
.6598	-.6221	.4707	1.0959		.9492	.2966	.7418	.6672						
.6997	-.5457	.4923	1.0592		1.0000	.1689	.7034	.7268						
.7493	-.4306	.5252	1.0648											
.8353	-.1879	.5901	.8576											
.8791	-.0773	.6315	.8375											
.9212	.0165	.6572	.7979											
1.0000	.1689	.7034	.7268											

TEST 122	PT	17.7760	PSI	CN	-2.948	CD1	.00720	CDCOR1	.00693
RUN 18	TT	197.0133	K	CM	-1.128	CD2	.00717	CDCOR2	.00686
POINT 3	RC	4.4783	MILLION	CC	.0075	CD3	.00774	CDCOR3	.00750
	MACH	.8025				CD4	.00656	CDCOR4	.00637
	ALPHA	.0264	DEG			CD5	.00525	CDCOR5	.00518

UPPER SURFACE					LOWER SURFACE					SPANWISE				
X/C	CP	P/L/PT	MLOC		X/C	CP	P/L/PT	MLOC		X/C	Y/8/2	CP	P/L/PT	MLOC
0.0000	1.1964	1.0074	0.0000		0.0000	1.1964	1.0074	0.0000		.0500	-.3375	-.1577	.6067	.8758
.0083	.3420	.7554	.6457		.0052	.1322	.6929	.7432		.3957	-.3375	-.5074	.5038	1.0401
.0097	.3345	.7526	.6501		.0098	.0876	.6859	.7539		.5008	-.3375	-.5742	.4869	1.0682
.0203	.0043	.6585	.7959		.0200	-.0648	.6746	.7712		.6048	-.3375	-.6725	.4557	1.1219
.0300	-.1155	.6215	.8528		.0500	-.0366	.6444	.8177		.7003	-.3375	-.5409	.4952	1.0544
.0400	-.1957	.5975	.8901		.0813	-.1028	.6261	.8457						
.0608	-.2794	.5743	.9264		.1199	-.1515	.6120	.8675						
.0800	-.3206	.5624	.9452		.1796	-.2253	.5886	.9040						
.1000	-.3602	.5488	.9667		.2397	-.2859	.5709	.9316						
.1997	-.4336	.5285	.9995		.2995	-.3556	.5514	.9626						
.2500	-.4534	.5238	1.0072		.3588	-.4313	.5302	.9967						
.2994	-.4804	.5146	1.0222		.4193	-.5053	.5072	1.0343						
.3402	-.4880	.5126	1.0254		.4793	-.5151	.5046	1.0386						
.3795	-.5036	.5064	1.0338		.5394	-.5787	.4832	.9758						
.4201	-.5164	.5034	1.0407		.5994	-.6370	.4636	.8807						
.4598	-.5452	.4952	1.0543		.6597	.0253	.6631	.7890						
.4996	-.5719	.4859	1.0699		.7203	.1767	.7067	.7218						
.5397	-.6101	.4740	1.0903		.7743	.2578	.7302	.6853						
.5795	-.6541	.4623	1.1104		.8394	.3154	.7479	.6575						
.6197	-.6904	.4509	1.1305		.8996	.3328	.7526	.6501						
.6598	-.7086	.4457	1.1394		.9492	.2988	.7428	.6656						
.6997	-.5668	.4865	1.0689		1.0000	.1562	.7003	.7317						
.7493	-.4265	.5283	.9999											
.8353	-.1766	.6020	.8831											
.8791	-.0690	.6333	.8347											
.9212	.0161	.6601	.7936											
1.0000	.1562	.7003	.7317											

TEST	122	PT	17.7733	PSI	CM	.4346	CD1	.00840	CDCOR1	.00800		
RUN	18	TT	196.9988	K	CM	-.1145	CD2	.00858	CDCOR2	.00825		
POINT	4	PC	4.4685	MILLION	CC	.0034	CD3	.00873	CDCOR3	.00835		
		MACH	.8001				CD4	.00721	CDCOR4	.00696		
		ALPHA	1.0064	DEG			CD5	.00565	CDCOR5	.00557		
		UPPER SURFACE			LOWER SURFACE			SPANWISE				
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC	X/C	Y/8/2	CP	P _L /PT	MLOC
0.0000	1.1231	.9860	.1423	0.0000	1.1231	.9860	.1423	.0500	-.3375	-.2972	.5692	.9343
.0083	.1323	.6940	.7413	.0052	.4379	.7841	.5996	.3957	-.3375	-.5964	.4806	1.0790
.0097	.1086	.6071	.7521	.0098	.3498	.7585	.6408	.5008	-.3375	-.6499	.4439	1.1077
.0203	-.2000	.5967	.8914	.0200	.2649	.7332	.6806	.6048	-.3375	-.7467	.4360	1.1568
.0300	-.3133	.5629	.9443	.0500	.1173	.6901	.7474	.7003	-.3375	-.5158	.5037	1.0402
.0400	-.3952	.5393	.9820	.0813	.0288	.6639	.7877					
.0608	-.4622	.5193	1.0144	.1199	-.0342	.6461	.8150					
.0800	-.4937	.5111	1.0279	.1796	-.1196	.6219	.8523					
.1000	-.5285	.5019	1.0431	.2397	-.1921	.5999	.8863					
.1997	-.5965	.4807	1.0788	.2995	-.2650	.5781	.9203					
.2500	-.5959	.4810	1.0783	.3588	-.3440	.5550	.9569					
.2994	-.5883	.4838	1.0735	.4193	-.4090	.5364	.9867					
.3402	-.5810	.4857	1.0702	.4793	-.4296	.5302	.9967					
.3795	-.5719	.4823	1.0761	.5394	-.3423	.5556	.9559					
.4201	-.6043	.4783	1.0828	.5994	-.1505	.6117	.8680					
.4598	-.6294	.4707	1.0958	.6507	.0448	.6690	.7799					
.4996	-.6451	.4665	1.1031	.7203	.1945	.7132	.7117					
.5397	-.6800	.4550	1.1230	.7743	.2766	.7367	.6752					
.5795	-.7219	.4440	1.1424	.8394	.3288	.7527	.6499					
.6197	-.7648	.4311	1.1657	.8996	.3422	.7565	.6439					
.6598	-.7762	.4272	1.1726	.9492	.2996	.7437	.6641					
.6997	-.5326	.4997	1.0468	1.0000	.1420	.6971	.7366					
.7493	-.4034	.5304	.9868									
.8353	-.1694	.6064	.8763									
.8791	-.0637	.6359	.8307									
.9212	.0178	.6615	.7914									
1.0000	.1420	.6971	.7366									

TEST	122	PT	17.7642	PSI	CM	-.5997	CD1	.01063	CDCOR1	.01010		
RUN	18	TT	197.2145	K	CM	-.1184	CD2	.00979	CDCOR2	.00925		
POINT	6	PC	4.4423	MILLION	CC	-.0040	CD3	.01064	CDCOR3	.01012		
		MACH	.7995				CD4	.00912	CDCOR4	.00865		
		ALPHA	2.0490	DEG			CD5	.00762	CDCOR5	.00723		
		UPPER SURFACE			LOWER SURFACE			SPANWISE				
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC	X/C	Y/8/2	CP	P _L /PT	MLOC
0.0000	.9803	.9436	.2889	0.0000	.9803	.9436	.2889	.0500	-.3375	-.4494	.5237	1.0074
.0083	-.1127	.6213	.8532	.0052	.6838	.8574	.4738	.3957	-.3375	-.7856	.4254	1.1759
.0097	-.1588	.6101	.8705	.0098	.5511	.8176	.5439	.5008	-.3375	-.8157	.4172	1.1909
.0203	-.4175	.5325	.9930	.0200	.4379	.7841	.5996	.6048	-.3375	-.8603	.4021	1.2193
.0300	-.5139	.5038	1.0400	.0500	.2567	.7320	.6825	.7003	-.3375	-.4378	.5284	.9997
.0400	-.6022	.4799	1.0801	.0813	.1494	.7000	.7322					
.0608	-.6636	.4611	1.1125	.1199	.0717	.6749	.7708					
.0800	-.6779	.4534	1.1259	.1796	-.0210	.6505	.8092					
.1000	-.7167	.4464	1.1383	.2397	-.1004	.6261	.8459					
.1997	-.7695	.4295	1.1686	.2995	-.1800	.6028	.8818					
.2500	-.7959	.4274	1.1815	.3598	-.2589	.5801	.9172					
.2994	-.8032	.4201	1.1857	.4193	-.3225	.5613	.9469					
.3402	-.8038	.4199	1.1860	.4793	-.3577	.5510	.9633					
.3795	-.8061	.4193	1.1871	.5394	-.2937	.5698	.9334					
.4201	-.7942	.4227	1.1808	.5994	-.1259	.6192	.8565					
.4598	-.7982	.4225	1.1813	.6507	.0666	.6763	.7687					
.4996	-.7819	.4260	1.1748	.7203	.2148	.7191	.7027					
.5397	-.8086	.4163	1.1926	.7743	.2916	.7407	.6689					
.5795	-.8370	.4078	1.2085	.8394	.3412	.7552	.6460					
.6197	-.8943	.4034	1.2167	.8996	.3487	.7578	.6418					
.6598	-.7469	.4353	1.1581	.9492	.3009	.7443	.6632					
.6997	-.4729	.5173	1.0178	1.0000	.1357	.6967	.7372					
.7493	-.3587	.5510	.9633									
.8353	-.1599	.6092	.8719									
.8791	-.0637	.6364	.8299									
.9212	.0149	.6609	.7922									
1.0000	.1357	.6967	.7372									

TEST	122	PT	17.7674	PSI	CM	.6696	CD1	.01274	CDCOR1	.01215		
RUN	18	TT	197.1761	K	CM	-.1241	CD2	.01198	CDCOR2	.01136		
POINT	7	PC	4.4409	MILLION	CC	-.0058	CD3	.01317	CDCOR3	.01256		
		MACH	.7992				CD4	.01202	CDCOR4	.01151		
		ALPHA	2.4807	DEG			CD5	.01102	CDCOR5	.01068		
		UPPER SURFACE			LOWER SURFACE			SPANWISE				
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC	X/C	Y/8/2	CP	P _L /PT	MLOC
0.0000	.9068	.9231	.3398	0.0000	.9068	.9231	.3398	.0500	-.3375	-.4930	.5107	1.0286
.0083	-.1934	.6008	.8849	.0052	.7536	.8779	.4353	.3957	-.3375	-.8436	.4069	1.2102
.0097	-.2496	.5834	.9121	.0098	.6200	.8384	.5080	.5008	-.3375	-.8938	.3951	1.2326
.0203	-.4981	.5098	1.0301	.0200	.4928	.8001	.5734	.6048	-.3375	-.9397	.3803	1.2612
.0300	-.5785	.4843	1.0727	.0500	.3040	.7451	.6619	.7003	-.3375	-.4913	.5222	1.0097
.0400	-.6616	.4611	1.1124	.0813	.1979	.7159	.7077					
.0608	-.7432	.4404	1.1488	.1199	.1139	.6908	.7463					
.0800	-.7665	.4329	1.1624	.1796	.0116	.6603	.7932					
.1000	-.7833	.4272	1.1727	.2397	-.0700	.6333	.8347					
.1997	-.8356	.4126	1.1996	.2995	-.1479	.6140	.8644					
.2500	-.8539	.4039	1.2158	.3598	-.2306	.5874	.9058					
.2994	-.8813	.3988	1.2256	.4193	-.2932	.5712	.9313					
.3402	-.8675	.4007	1.2218	.4793	-.3333	.5578	.9524					
.3795	-.8784	.4004	1.2224	.5394	-.2747	.5772	.9218					
.4201	-.8923	.3993	1.2245	.5994	-.1173	.6233	.8501					
.4598	-.8780	.4003	1.2226	.6507	.0700	.6780	.7660					
.4996	-.8624	.3993	1.2241	.7203	.2149	.7176	.7050					
.5397	-.8716	.3995	1.2240	.7743	.2947	.7425	.6661					
.5795	-.9120	.3874	1.2474	.8394	.3404	.7558	.6450					
.6197	-.9359	.3813	1.2593	.8996	.3469	.7581	.6413					
.6598	-.8405	.4057	1.2123	.9492	.3027	.7432	.6649					
.6997	-.6499	.5234	1.0677	1.0000	.1215	.6921	.7443					
.7493	-.3537	.5505	.9641									
.8353	-.1542	.6092	.8720									
.8791	-.0709	.6346	.8328									
.9212	-.0612	.6541	.8027									
1.0000	.1215	.6921	.7443									

TEST	122	PT	17.7807	PSI	CN	.7188	CD1	.01680	CDCOR1	.01603
RUN	18	TT	197.1818	K	CM	-.1278	CD2	.01796	CDCOR2	.01702
POINT	8	RC	4.4536	MILLION	CC	-.0065	CD3	.02175	CDCOR3	.02098
		MACH	.8032				CD4	.01895	CDCOR4	.01802
		ALPHA	2.8693	DEG			CD5	.01797	CDCOR5	.01730

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _s L/PT	MLOC	X/C	CP	P _s L/PT	MLOC	X/C	Y/B/2	CP	P _s L/PT	MLOC
0.0000	.8429	.9036	.3831	0.0000	.8429	.9036	.3831	.0500	-.3375	-.5304	.4977	1.0502
.0083	-.2512	.5817	.9148	.0083	.8096	.8932	.4048	.3957	-.3375	-.8958	.3892	1.2439
.0097	-.3218	.5592	.9502	.0098	.6785	.8541	.4798	.5008	-.3375	-.9356	.3765	1.2687
.0203	-.5523	.4903	1.0626	.0200	.5474	.8161	.5464	.6048	-.3375	-1.0190	.3548	1.3126
.0300	-.6386	.4666	1.1029	.0300	.3412	.7543	.6475	.7003	-.3375	-.4608	.5163	1.0193
.0400	-.7027	.4455	1.1398	.0813	.2287	.7200	.7012					
.0608	-.7707	.4236	1.1792	.1199	.1438	.6968	.7370					
.0800	-.8093	.4156	1.1939	.1796	.0411	.6670	.7830					
.1000	-.8342	.4090	1.2063	.2397	-.0468	.6408	.8231					
.1197	-.8844	.3957	1.2314	.2995	-.1266	.6185	.8574					
.2500	-.9055	.3882	1.2458	.3588	-.2092	.5934	.8963					
.2994	-.9327	.3802	1.2614	.4193	-.2784	.5730	.9284					
.3402	-.9317	.3786	1.2646	.4793	-.3233	.5583	.9516					
.3795	-.9287	.3789	1.2639	.5394	-.2758	.5720	.9300					
.4201	-.9353	.3785	1.2648	.5994	-.1207	.6188	.8570					
.4598	-.9475	.3738	1.2741	.6507	.0708	.6747	.7711					
.4996	-.9465	.3770	1.2677	.7203	.2173	.7195	.7020					
.5397	-.9387	.3745	1.2727	.7743	.2915	.7390	.6716					
.5795	-.9622	.3700	1.2817	.8394	.3404	.7546	.6470					
.6197	-1.0013	.3576	1.3067	.8996	.3421	.7547	.6667					
.6598	-.7624	.4272	1.1726	.9492	.2907	.7390	.6716					
.6997	-.4489	.5212	1.0114	1.0000	.0976	.6826	.7590					
.7493	-.3696	.5435	.9753									
.8353	-.1824	.6008	.8849									
.8791	-.1031	.6238	.8493									
.9212	-.0117	.6525	.8051									
1.0000	.0976	.6826	.7590									

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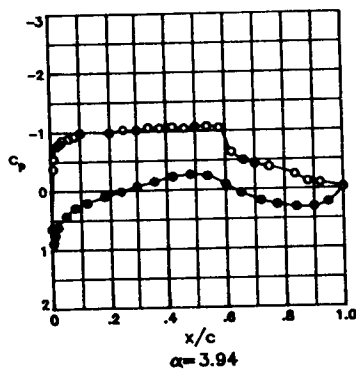
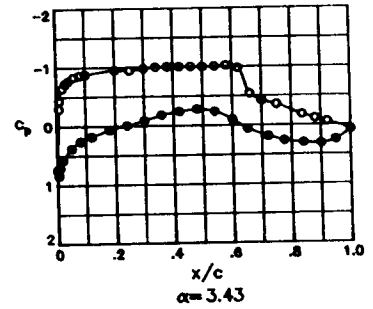
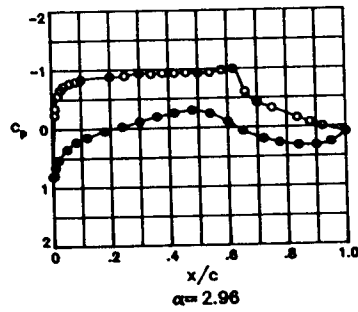
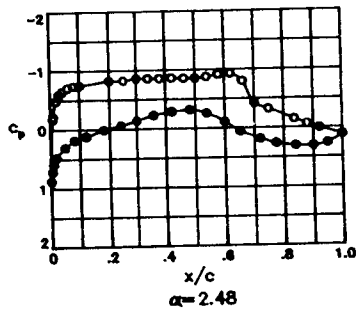
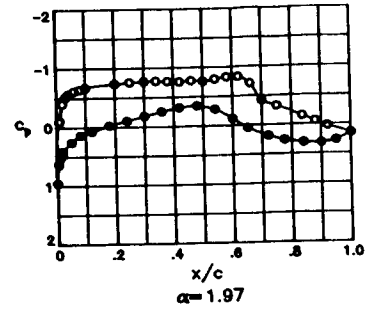
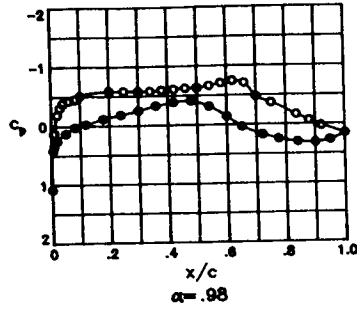
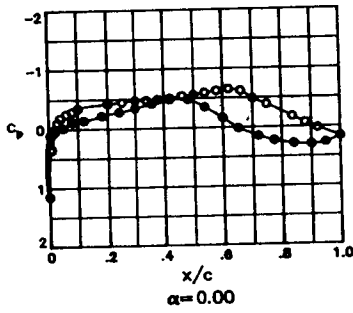
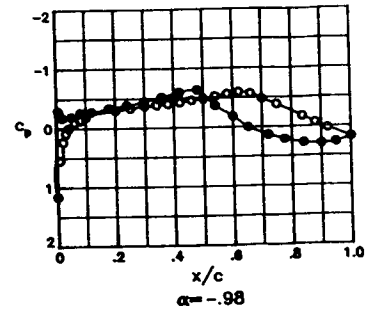
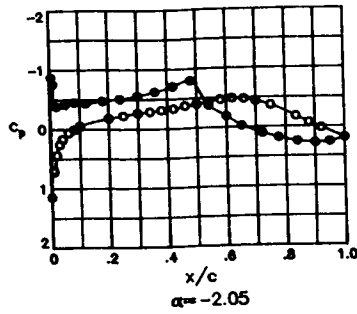
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RUN	18	TT	197.4093	K	CM	-.1348	CD2	.02188	CDCOR2	.02097
POINT	9	RC	4.4349	MILLION	CC	-.0069	CD3	.02784	CDCOR3	.02640
		MACH	.8002				CD4	.02507	CDCOR4	.02417
		ALPHA	3.4567	DEG			CD5	.02220	CDCOR5	.02118

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _s L/PT	MLOC	X/C	CP	P _s L/PT	MLOC	X/C	Y/B/2	CP	P _s L/PT	MLOC
0.0000	.7741	.8841	.4230	0.0000	.7741	.8841	.4230	.0500	-.3375	-.5754	.4854	1.0709
.0083	-.3185	.5637	.9431	.0052	.8767	.9139	.3609	.3957	-.3375	-.9490	.3747	1.2722
.0097	-.4517	.5237	1.0072	.0098	.7310	.8700	.4503	.5008	-.3375	-.9825	.3648	1.2921
.0203	-.6257	.4696	1.0978	.0200	.5937	.8303	.5221	.6048	-.3375	-1.0619	.3426	1.3380
.0300	-.7036	.4486	1.1343	.0300	.3933	.7728	.6178	.7003	-.3375	-.4607	.5212	1.0114
.0400	-.7922	.4257	1.1754	.0813	.2726	.7363	.6758					
.0608	-.8535	.4053	1.2131	.1199	.1816	.7086	.7188					
.0800	-.8752	.3974	1.2282	.1796	.0742	.6770	.7676					
.1000	-.8935	.3919	1.2386	.2397	-.0150	.6507	.8080					
.1197	-.9362	.3794	1.2630	.2995	-.0963	.6268	.8448					
.2500	-.9697	.3732	1.2752	.3588	-.1843	.6034	.8809					
.2994	-.9825	.3650	1.2898	.4193	-.2593	.5749	.9191					
.3402	-.9948	.3627	1.2963	.4793	-.2986	.5676	.9369					
.3795	-1.0084	.3597	1.3025	.5394	-.2579	.5803	.9170					
.4201	-.9906	.3627	1.2964	.5994	-.1147	.6209	.8538					
.4598	-1.0129	.3588	1.3043	.6507	.0720	.6777	.7665					
.4996	-1.0257	.3571	1.3078	.7203	.2178	.7213	.6991					
.5397	-1.0276	.3561	1.3099	.7743	.2942	.7435	.6645					
.5795	-1.0227	.3535	1.3152	.8394	.3335	.7532	.6492					
.6197	-.9945	.3642	1.2933	.8996	.3353	.7548	.6666					
.6598	-.5645	.5212	1.0630	.9492	.2898	.7405	.6693					
.6997	-.4492	.5228	1.0088	1.0000	.0495	.6706	.7774					
.7493	-.3881	.5404	.9802									
.8353	-.2255	.5891	.9031									
.8791	-.1315	.6178	.8587									
.9212	-.0050	.6267	.8449									
1.0000	.0495	.6706	.7774									

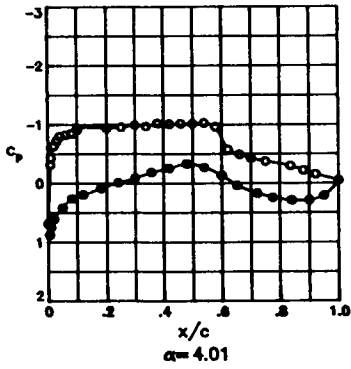
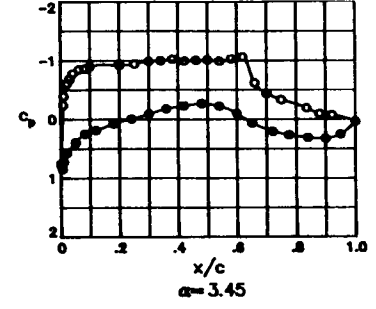
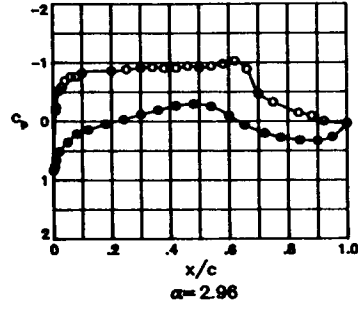
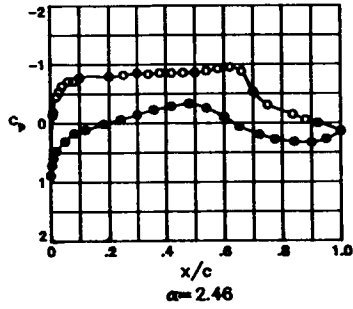
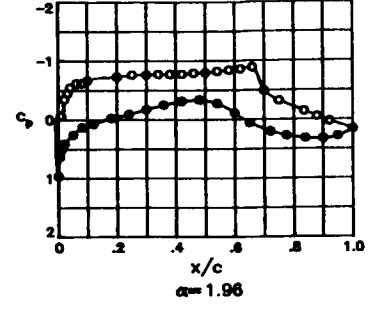
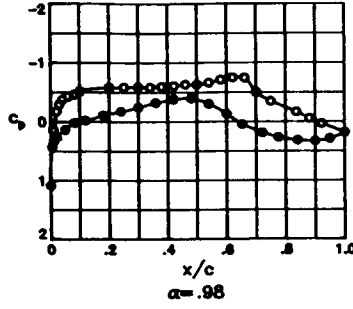
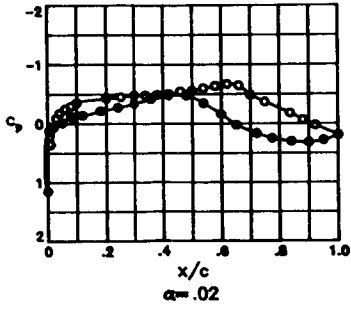
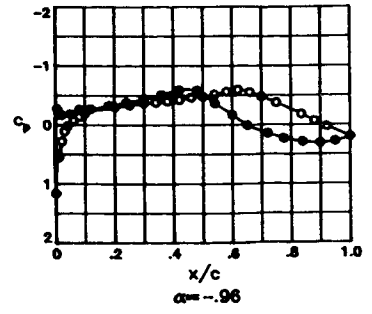
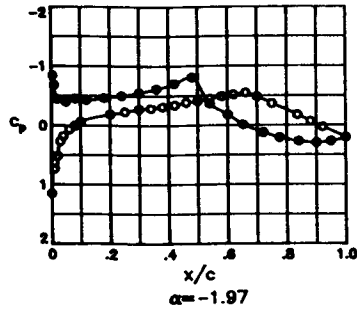
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RUN	18	TT	197.4787	K	CM	-.1383	CD2	.03082	CDCOR2	.02966
POINT	10	RC	4.4404	MILLION	CC	-.0057	CD3	.04363	CDCOR3	.04218
		MACH	.8027				CD4	.03788	CDCOR4	.03679
		ALPHA	3.9486	DEG			CD5	.02947	CDCOR5	.02876

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _s L/PT	MLOC	X/C	CP	P _s L/PT	MLOC	X/C	Y/B/2	CP	P _s L/PT	MLOC
0.0000	.7162	.8656	.4586	0.0000	.7162	.8656	.4586	.0500	-.3375	-.6152	.4742	1.0899
.0083	-.3650	.5464	.9707	.0052	.9169	.9251	.3351	.3957	-.3375	-.9961	.3594	1.3030
.0097	-.5303	.4987	1.0484	.0098	.7748	.8827	.4257	.5008	-.3375	-1.0183	.3325	1.3173
.0203	-.6954	.4484	1.1346	.0200	.6375	.8439	.4982	.6048	-.3375	-.6229	.4672	1.1019
.0300	-.7722	.4304	1.1668	.0300	.4233	.7800	.6062	.7003	-.3375	-.4581	.5162	1.0163
.0400	-.8299	.4112	1.2022	.0813	.3041	.7449	.6622					
.0608	-.8963	.3916	1.2392	.1199	.2060	.7155	.7083					
.0800	-.9127	.3856	1.2508	.1796	.0959	.6822	.7595					
.1000	-.9320	.3787	1.2644	.2397	.0026	.6556	.8004					
.1197	-.9840	.3668	1.2880	.2995	-.0754	.6339	.8338					
.2500	-1.0070	.3605	1.3009	.3588	-.1685	.6068	.8756					
.2994	-1.0418	.3532	1.3157	.4193	-.2380	.5884	.9042					
.3402	-1.0299	.3496	1.3232	.4793	-.3038	.5641	.9424					
.3795	-1.0464	.3438	1.3356	.5394	-.2632	.5754	.9246					
.4201	-1.0535	.3444	1.3341	.5994	-.1145	.6211	.8534					
.4598	-1.0561	.3449	1.3331	.6507	.0710	.6765	.7683					
.4996	-1.0594	.3406	1.3423	.7203	.2076	.7150	.7089					
.5397	-1.0617	.3374	1.3491	.7743	.2849	.7394	.6708					
.5795	-.9487	.3754	1.2709	.8394	.3239	.7504	.6536					
.6197	-.5865	.4603	1.0795	.8996	.3204	.7483	.6569					
.6598	-.4896	.5111	1.0279	.9492	.2552	.7304	.6890					
.6997	-.4529	.5215	1.0108	1.0000	-.0665	.6325	.8359					
.7493	-.4248	.5288	.9990									
.8353	-.3240	.5572	.9535									
.8791	-.2364	.5849	.9097									
.9212	-.2014	.5954	.8934									
1.0000	-.0465	.6325	.8359									

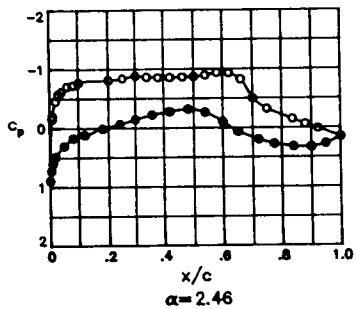
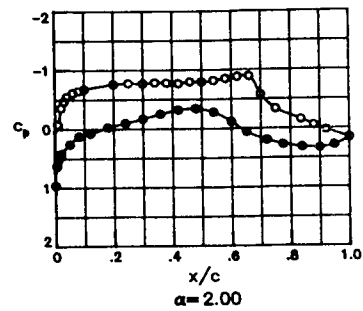
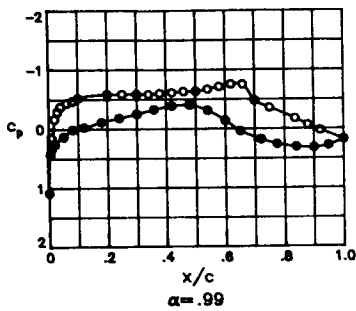
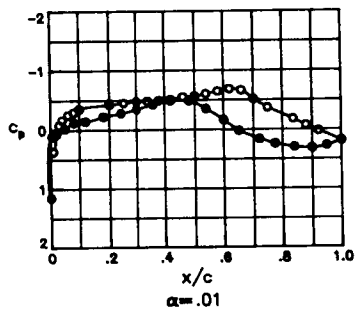
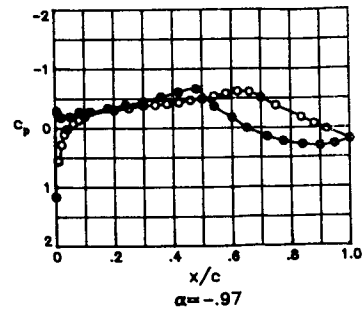
TEST 122
 RUN 26
 MACH .807
 R 7.7×10^6



TEST 122
 RUN 33
 MACH .807
 R 14.0×10^6



TEST 122
 RUN 35
 MACH .807
 R 14.0×10^6



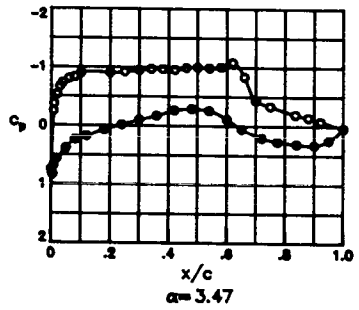
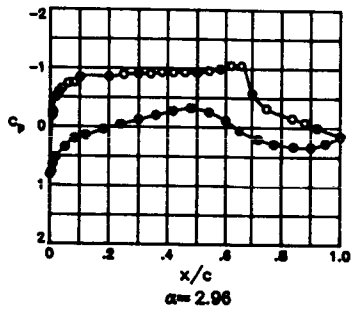
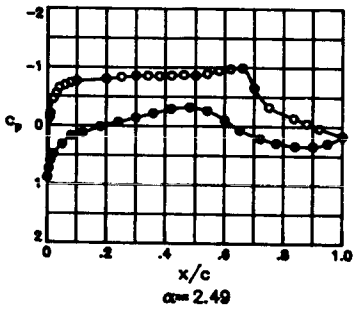
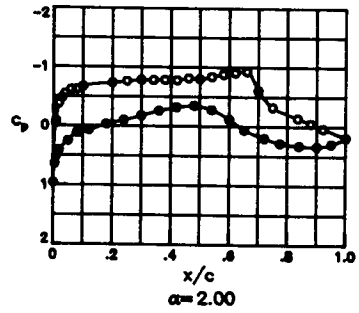
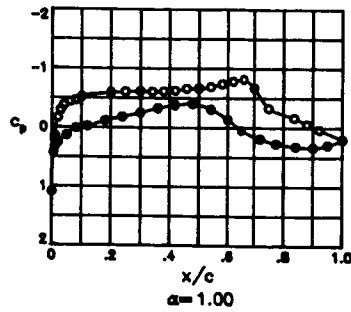
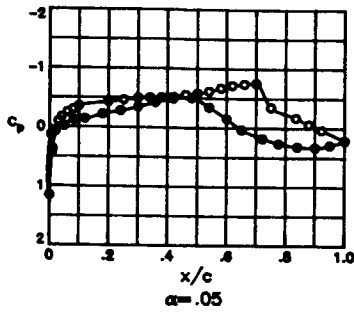
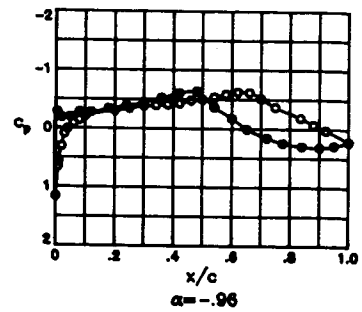
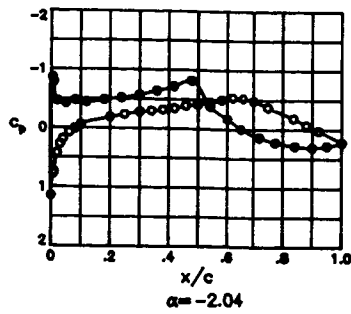
TEST 122	PT	20.8269	PSI	CN	.5908	CD1	.01252	CDCOR1	.C1204
RUN 35	TT	99.9689	K	CM	-1.192	CD2	.01229	CDCOR2	.01179
POINT 4	RC	13.9660	MILLION	CC	-0.0023	CD3	.01231	CDCOR3	.01182
	MACH	.8031				CD4	.01162	CDCOR4	.01129
	ALPHA	2.0004	DEG			CD5	.01128	CDCOR5	.01102

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P,L/PT	MLOC	X/C	CP	P,L/PT	MLOC	X/C	Y/B/2	CP	P,L/PT	MLOC
0.0000	.9601	.9379	.3045	0.0000	.9601	.9379	.3045	.0500	-.3375	-.5083	.5084	1.0338
.0083	-.0815	.6316	.8387	.0052	.6365	.8412	.5040	.3957	-.3375	-.7712	.4278	1.1730
.0097	-.0692	.6324	.8375	.0098	.5240	.8083	.5607	.5008	-.3375	-.8081	.4192	1.1887
.0203	-.3594	.5473	.9708	.0200	.4202	.7793	.6086	.6048	-.3375	-.8658	.3976	1.2291
.0300	-.4609	.5177	1.0187	.0500	.2596	.7307	.6859	.7003	-.3375	-.5708	.4819	1.0782
.0400	-.5443	.4923	1.0607	.0813	.1256	.6923	.7454					
.0608	-.6145	.4746	1.0908	.1199	.0765	.6788	.7662					
.0800	-.6465	.4666	1.1145	.1796	-.0256	.6469	.8153					
.1000	-.6749	.4555	1.1238	.2397	-.0962	.6265	.8466					
.1997	-.7578	.4371	1.1563	.2995	-.1695	.6089	.8738					
.2500	-.7742	.4297	1.1696	.3588	-.2493	.5837	.9132					
.2994	-.7696	.4242	1.1797	.4193	-.3207	.5572	.9549					
.3402	-.7434	.4274	1.1737	.4793	-.3410	.5371	.9951					
.3795	-.7781	.4250	1.1782	.5394	-.2781	.5224	.9309					
.4201	-.7704	.4214	1.1847	.5994	-.1173	.6157	.8634					
.4598	-.7449	.4195	1.1883	.6507	.0605	.6719	.7768					
.4996	-.7945	.4225	1.1826	.7203	.1992	.7145	.7111					
.5397	-.8134	.4189	1.1894	.7743	.2769	.7375	.6752					
.5795	-.8460	.4061	1.2132	.8394	.3240	.7537	.6497					
.6197	-.8765	.3970	1.2304	.8996	.3311	.7543	.6487					
.6598	-.8852	.3863	1.2508	.9492	.2729	.7394	.6722					
.6997	-.8734	.4861	1.0712	1.0000	.1615	.7001	.7333					
.7493	-.3301	.5601	.9503									
.8353	-.1483	.6139	.8661									
.8791	-.0486	.6398	.8262									
.9212	.0300	.6662	.7855									
1.0000	.1615	.7001	.7333									

TEST 122	PT	20.8318	PSI	CN	.6544	CD1	.01484	CDCOR1	.01421
RUN 35	TT	100.0601	K	CM	-1.202	CD2	.01451	CDCOR2	.01386
POINT 5	RC	13.9180	MILLION	CC	-0.0055	CD3	.01417	CDCOR3	.01354
	MACH	.8007				CD4	.01509	CDCOR4	.01464
	ALPHA	2.4600	DEG			CD5	.01472	CDCOR5	.01434

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P,L/PT	MLOC	X/C	CP	P,L/PT	MLOC	X/C	Y/B/2	CP	P,L/PT	MLOC
0.0000	.8905	.9172	.3541	0.0000	.8905	.9172	.3541	.0500	-.3375	-.5666	.4907	1.0635
.0083	-.1551	.6093	.8731	.0052	.7254	.8702	.4509	.3957	-.3375	-.8449	.4097	1.2063
.0097	-.2019	.5991	.8890	.0098	.6030	.8347	.5153	.5008	-.3375	-.9034	.3932	1.2377
.0203	-.4593	.5247	1.0071	.0200	.4836	.8002	.5743	.6048	-.3375	-.9361	.3751	1.2730
.0300	-.5694	.4931	1.0593	.0500	.3027	.7426	.6672	.7003	-.3375	-.4630	.5223	1.0111
.0400	-.6142	.4714	1.0961	.0813	.1717	.7075	.7219					
.0608	-.7061	.4504	1.1327	.1199	.1163	.6911	.7472					
.0800	-.7219	.4455	1.1414	.1796	.0106	.6612	.7932					
.1000	-.7683	.4335	1.1628	.2397	-.0669	.6359	.8322					
.1997	-.8096	.4180	1.1909	.2995	-.1447	.6134	.8669					
.2500	-.8382	.4098	1.2063	.3588	-.2213	.5910	.9017					
.2994	-.8798	.4043	1.2164	.4193	-.2758	.5800	.9188					
.3402	-.8602	.4067	1.2120	.4793	-.3176	.5653	.9421					
.3795	-.8524	.4061	1.2132	.5394	-.2621	.5794	.9199					
.4201	-.8547	.4034	1.2183	.5994	-.1032	.6246	.8495					
.4598	-.8627	.4031	1.2189	.6507	.0650	.6754	.7714					
.4996	-.8686	.4008	1.2231	.7203	.2010	.7150	.7102					
.5397	-.8912	.3945	1.2350	.7743	.2769	.7375	.6752					
.5795	-.9331	.3871	1.2494	.8394	.3240	.7537	.6497					
.6197	-.9268	.3859	1.2517	.8996	.3311	.7543	.6487					
.6598	-.8204	.4212	1.1852	.9492	.2729	.7394	.6722					
.6997	-.4975	.5087	1.0334	1.0000	.1567	.7045	.7266					
.7493	-.3187	.5645	.9434									
.8353	-.1466	.6110	.8705									
.8791	-.0591	.6369	.8307									
.9212	.0125	.6572	.7994									
1.0000	.1567	.7045	.7266									

TEST 122
 RUN 40
 MACH .807
 R 30.0×10^6



TEST 122 PT 51.8151 PSI CN .4467
 RUN 40 TT 109.8245 K CM -1166
 POINT 4 RC 30.2690 MILLION CC .0032
 MACH .#009
 ALPHA 1.0000 DEG

CD1 .00831 CDCOR1 .00802
 CD2 .00830 CDCOR2 .00792
 CD3 .01841 CDCOR3 .01807
 CD4 .00795 CDCOR4 .00774
 CD5 .00765 CDCOR5 .00753

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC
0.0000	1.0948	.9775	.1811	0.0000	1.0948	.9775	.1811
.0083	.1984	.7142	.7123	.0052	.4281	.7815	.6056
.0097	.1452	.6984	.7368	.0098	.3289	.7525	.6522
.0203	-.1810	.6086	.8753	.0200	.2580	.7315	.6852
.0300	-.2949	.5691	.9372	.0500	.1341	.6959	.7406
.0400	-.3729	.5472	.9719	.0813	.0115	.6592	.7972
.0608	-.4319	.5289	1.0115	.1199	-.0219	.6492	.8126
.0800	-.4611	.5201	1.0154	.1796	-.1166	.6221	.8543
.1000	-.5173	.5038	1.0427	.2397	-.1796	.6038	.8827
.1997	-.5903	.4838	1.0762	.2995	-.2491	.5838	.9139
.2500	-.6606	.4801	1.0826	.3588	-.3271	.5616	.9490
.2994	-.6110	.4795	1.0835	.4193	-.3811	.5467	.9727
.3402	-.6054	.4791	1.0842	.4793	-.4041	.5382	.9866
.3795	-.6134	.4778	1.0864	.5394	-.3110	.5663	.9415
.4201	-.6279	.4747	1.0918	.5994	-.1321	.5196	.8583
.4598	-.6590	.4443	1.1096	.6507	.0483	.6714	.7784
.4996	-.6841	.4635	1.1110	.7203	.1943	.7146	.7117
.5397	-.6894	.4569	1.1225	.7743	.2799	.7389	.6737
.5795	-.7363	.4428	1.1474	.8394	.3244	.7528	.6517
.6197	-.7808	.4297	1.1709	.8996	.3401	.7574	.6444
.6598	-.8157	.4194	1.1897	.9492	.2969	.7447	.6645
.6997	-.6704	.4616	1.1143	1.0000	.2031	.7159	.7097
.7493	-.3268	.5614	.9493				
.8353	-.1603	.6104	.8726				
.8791	-.0533	.6416	.8244				
.9212	.0378	.6679	.7839				
1.0000	.2031	.7159	.7097				

ORIGINAL PAGE IS
 OF POOR QUALITY

TEST 122 PT 51.8211 PSI CN .5929
 RUN 40 TT 110.4194 K CM -1224
 POINT 5 RC 29.9240 MILLION CC -0.0027
 MACH .#013
 ALPHA 2.0000 DEG

CD1 .01184 CDCOR1 .01144
 CD2 .01177 CDCOR2 .01123
 CD3 .02512 CDCOR3 .02436
 CD4 .01185 CDCOR4 .01136
 CD5 .01125 CDCOR5 .01059

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC
0.0000	.9635	.9386	.3030	0.0000	.9635	.9386	.3030
.0083	-.0440	.6420	.8236	.0052	.6469	.8464	.4952
.0097	-.0790	.6336	.8366	.0098	.5175	.8085	.5609
.0203	-.3809	.5452	.9752	.0200	.4135	.7768	.6132
.0300	-.4695	.5149	1.0210	.0500	.2554	.7293	.6886
.0400	-.5423	.4940	1.0591	.0813	.1183	.6903	.7491
.0608	-.6152	.4747	1.0916	.1199	-.0735	.6450	.8100
.0800	-.6234	.4707	1.0985	.1796	-.0305	.6450	.8100
.1000	-.6662	.4575	1.1215	.2397	-.0981	.6268	.8471
.1997	-.7305	.4398	1.1527	.2995	-.1760	.6031	.8838
.2500	-.7591	.4307	1.1690	.3588	-.2591	.5781	.9229
.2994	-.7790	.4254	1.1787	.4193	-.3201	.5605	.9507
.3402	-.7761	.4265	1.1767	.4793	-.3465	.5520	.9627
.3795	-.7809	.4254	1.1787	.5394	-.2785	.5732	.9305
.4201	-.7753	.4258	1.1779	.5994	-.1142	.5207	.8565
.4598	-.8109	.4208	1.1872	.6507	.0639	.6765	.7704
.4996	-.7974	.4162	1.1856	.7203	.2038	.7127	.7146
.5397	-.8358	.4131	1.2013	.7743	.2849	.7410	.6703
.5795	-.8777	.3935	1.2382	.8394	.3293	.7504	.6554
.6197	-.9036	.3920	1.2412	.8996	.3416	.7570	.6450
.6598	-.9274	.3826	1.2594	.9492	.2975	.7429	.6674
.6997	-.6026	.4831	1.0774	1.0000	.1896	.7142	.7122
.7493	-.3223	.5620	.9484				
.8353	-.1372	.6168	.8625				
.8791	-.0468	.6421	.8188				
.9212	.0370	.6694	.7813				
1.0000	.1896	.7142	.7122				

TEST 122 PT 51.8174 PSI CN .6686
 RUN 40 TT 110.3739 K CM -1270
 POINT 6 RC 29.9180 MILLION CC -0.0047
 MACH .#010
 ALPHA 2.4900 DEG

CD1 .01594 CDCOR1 .01553
 CD2 .01536 CDCOR2 .01481
 CD3 .02963 CDCOR3 .02920
 CD4 .01596 CDCOR4 .01564
 CD5 .01532 CDCOR5 .01503

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC
0.0000	.8854	.9171	.3545	0.0000	.8854	.9171	.3545
.0043	-.1409	.6177	.8611	.0052	.7326	.8726	.4466
.0097	-.2034	.5996	.8892	.0098	.5942	.8307	.5220
.0203	-.4982	.5219	1.0124	.0200	.4805	.7975	.5792
.0300	-.5497	.4954	1.0566	.0500	.3184	.7521	.6528
.0400	-.6463	.4710	1.0981	.0813	.1687	.7069	.7250
.0608	-.6942	.4528	1.1296	.1199	.1202	.6947	.7425
.0800	-.7345	.4457	1.1421	.1796	.0124	.6611	.7942
.1000	-.7596	.4351	1.1611	.2397	-.0682	.6343	.8355
.1997	-.8007	.4230	1.1831	.2995	-.1431	.6155	.8645
.2500	-.8329	.4136	1.2005	.3588	-.2198	.5931	.8994
.2994	-.8541	.4036	1.2192	.4193	-.2895	.5697	.9361
.3402	-.8566	.4072	1.2123	.4793	-.3183	.5647	.9441
.3795	-.8515	.4075	1.2117	.5394	-.2592	.5811	.9182
.4201	-.8577	.4046	1.2172	.5994	-.1034	.6254	.8493
.4598	-.8672	.4009	1.2243	.6507	.0677	.6756	.7719
.4996	-.8570	.4004	1.2252	.7203	.2096	.7154	.7103
.5397	-.8906	.3943	1.2367	.7743	.2866	.7401	.6717
.5795	-.9425	.3857	1.2534	.8394	.3364	.7579	.6436
.6197	-.9737	.3714	1.2816	.8996	.3427	.7573	.6446
.6598	-.9909	.3623	1.2998	.9492	.2896	.7398	.6723
.6997	-.6479	.4684	1.0132	1.0000	.1665	.7065	.7242
.7493	-.3257	.5638	.9454				
.8353	-.1387	.6150	.8653				
.8791	-.0487	.6419	.8237				
.9212	.0351	.6675	.7843				
1.0000	.1665	.7065	.7242				

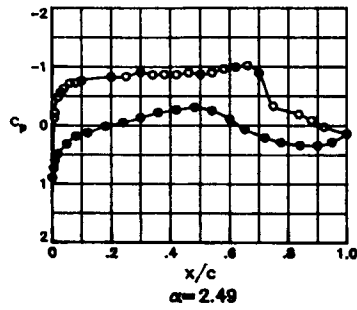
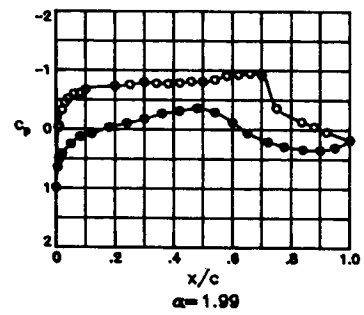
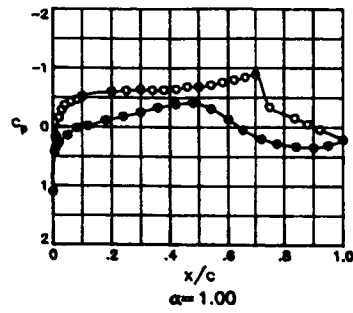
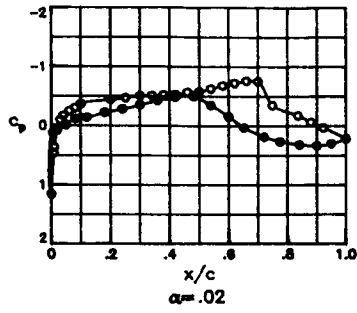
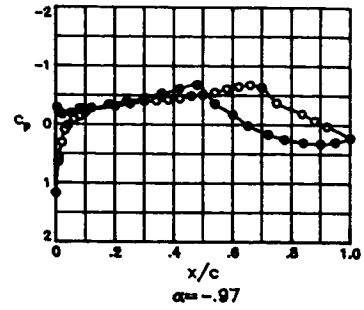
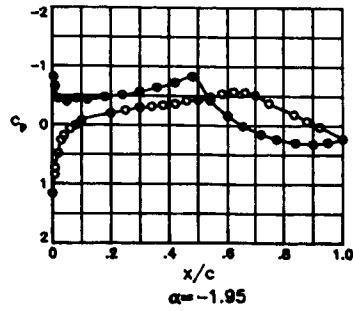
TEST 122	PT 51.8330	PSI	CN .7179	CD1 .01918	CDCOR1 .01812
RUN 40	TT 110.2149	K	CM -.1310	CD2 .02067	CDCOR2 .01972
POINT R	PC 29.9670	MILLION	CC -.0055	CD3 .03984	CDCOR3 .03886
	MACH .8011			CD4 .02063	CDCOR4 .01984
	ALPHA 2.9600	DEG		CD5 .02139	CDCOR5 .02080

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC	X/C	Y/R/2	CP	P _L /PT	MLOC
0.0000	.8158	.8964	.3994	0.0000	.8158	.8964	.3994	.0503	-.3375	-.5940	.4805	1.0818
.0063	-.1872	.8031	.8938	.0052	-.7760	.8833	.4259	.3957	-.3375	-.8691	.3947	1.2359
.0197	-.2475	.5818	.9171	.0098	-.6376	.8426	.5021	.5008	-.3375	-.9314	.3792	1.2461
.0203	-.5075	.9054	1.0401	.0200	.5210	.8067	.5640	.6048	-.3375	-1.0153	.3622	1.3002
.0300	-.5910	.4774	1.0871	.0500	.3464	.7542	.6494	.7003	-.3375	-.5620	.4931	1.0606
.0400	-.6493	.4588	1.1193	.0813	.1976	.7133	.7136					
.0600	-.7392	.4378	1.1564	.1199	.1451	.6964	.7399					
.0800	-.7510	.4317	1.1673	.1796	.0430	.6723	.7770					
.1000	-.8374	.4160	1.1960	.2397	-.0433	.6439	.8207					
.1997	-.8505	.4057	1.2148	.2995	-.1221	.6199	.8577					
.2500	-.8882	.3976	1.2305	.3588	-.2005	.5988	.8904					
.2994	-.9043	.3878	1.2494	.4193	-.2693	.5750	.9278					
.3402	-.9025	.3778	1.2493	.4793	-.3208	.5594	.9524					
.3795	-.9282	.3693	1.2464	.5394	-.2494	.5873	.9085					
.4201	-.9262	.3598	1.2454	.5994	-.1095	.6277	.8657					
.4598	-.9320	.3789	1.2667	.6507	.0608	.6719	.7777					
.4996	-.9239	.3844	1.2559	.7203	.2051	.7161	.7093					
.5397	-.9552	.3794	1.2657	.7743	.2876	.7424	.6682					
.5795	-.9859	.3697	1.2845	.8394	.3299	.7545	.6490					
.6197	-1.0487	.3572	1.3103	.8996	.3387	.7596	.6409					
.6598	-1.0470	.3528	1.3194	.9492	.2742	.7385	.6742					
.6997	-.5766	.4615	1.0801	1.0000	.1533	.7049	.7267					
.7493	-.3202	.5605	.9309									
.8393	-.1524	.6123	.8694									
.8791	-.0797	.6315	.8397									
.9212	.0071	.6585	.7983									
1.0000	.1533	.7649	.7267									

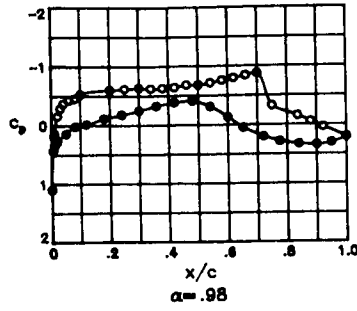
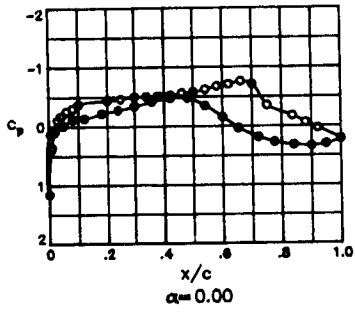
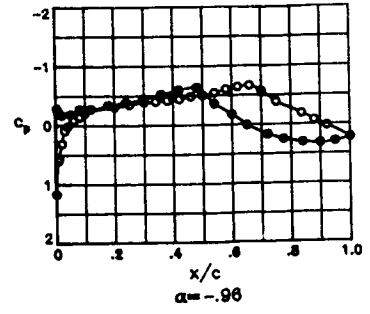
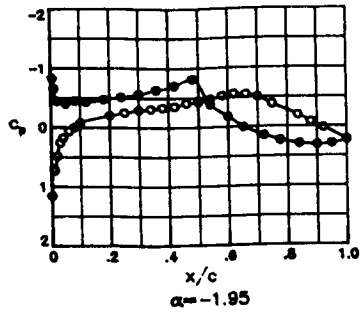
TEST 122	PT 51.8236	PSI	CN .7563	CD1 .02535	CDCOR1 .02496
RUN 40	TT 110.1048	K	CM -.1312	CD2 .02822	CDCOR2 .02780
POINT R	PC 30.1020	MILLION	CC -.0057	CD3 .06527	CDCOR3 .06482
	MACH .8053			CD4 .03155	CDCOR4 .03119
	ALPHA 3.4661	DEG		CD5 .03274	CDCOR5 .03251

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC	X/C	Y/R/2	CP	P _L /PT	MLOC
0.0000	.7476	.8765	.4391	0.0000	.7476	.8765	.4391	.0503	-.3375	-.6213	.4684	1.1024
.0063	-.2459	.5861	.9104	.0052	.8392	.9033	.3848	.3957	-.3375	-.9487	.3808	1.2630
.0197	-.3693	.5901	.9673	.0098	.6834	.8539	.4815	.5008	-.3375	-.9457	.3652	1.2939
.0203	-.5292	.4436	1.0596	.0200	.5645	.8211	.5396	.6048	-.3375	-1.0850	.3428	1.3403
.0300	-.6611	.4062	1.1167	.0500	.3890	.7689	.6260	.7003	-.3375	-.4592	.5223	1.0123
.0400	-.7316	.4384	1.1552	.0813	.2363	.7250	.6954					
.0600	-.8030	.4195	1.1895	.1199	.1793	.7093	.7198					
.0800	-.8296	.4136	1.2005	.1796	.0718	.6802	.7648					
.1000	-.8934	.3989	1.2279	.2397	-.0145	.6567	.8009					
.1997	-.8999	.3914	1.2474	.2995	-.1007	.6262	.8480					
.2500	-.9170	.3833	1.2581	.3588	-.1689	.6040	.8823					
.2994	-.9463	.3743	1.2758	.4193	-.2639	.5758	.9265					
.3402	-.9472	.3723	1.2798	.4793	-.2884	.5715	.9332					
.3795	-.9655	.3694	1.2855	.5394	-.2593	.5777	.9236					
.4201	-.9543	.3723	1.2798	.5994	-.1144	.6202	.8573					
.4598	-.9883	.3644	1.2956	.6507	.0543	.6712	.7787					
.4996	-.9466	.3636	1.2973	.7203	.2034	.7159	.7096					
.5397	-.9862	.3605	1.3036	.7743	.2686	.7320	.6845					
.5795	-1.0032	.3516	1.3218	.8394	.3104	.7426	.6679					
.6197	-1.0793	.3453	1.3349	.8996	.3283	.7552	.6478					
.6598	-.8343	.4071	1.2125	.9492	.2465	.7263	.6933					
.6997	-.4347	.5229	1.0112	1.0000	.0397	.6662	.7863					
.7493	-.3348	.5350	.9595									
.8393	-.1341	.5985	.8910									
.8791	-.1435	.6115	.8707									
.9212	-.0774	.6319	.8391									
1.0000	.0397	.6662	.7863									

TEST 122
RUN 49
MACH .807
R 45.0×10^6



TEST 122
RUN 48
MACH .807
R 50.0×10^6



TEST	122	PT	79.4807	PSI		CN	.4679	CD1	.00804	CDCDR1	.00726
RUN	48	TT	104.3018	K		CM	-.1232	CD2	.00805	CDCDR2	.00776
POINT	4	RC	50.0870	MILLION		CC	.0038	CD3	.01376	CDCDR3	.01345
		MAC+	.8022	DEG				CD4	.00781	CDCDR4	.00719
		ALPHA	.9800					CD5	.00733	CDCDR5	.00704

UPPER SURFACE					LOWER SURFACE					SPANWISE				
X/C	CP	P/L/PT	MLOC		X/C	CP	P/L/PT	MLOC		X/C	Y/B/2	CP	P/L/PT	MLOC
0.0000	1.0975	.9782	.1785		0.0000	1.0975	.9782	.1785		.0503	-.3375	-.2714	.5774	.9264
.0043	.1967	.7145	.7137		.0052	.4326	.7836	.6039		.3957	-.3375	-.6404	.4709	1.1008
.0097	.1461	.6995	.7368		.0098	.3451	.7576	.6459		.5008	-.3375	-.6851	.4569	1.1251
.0203	-.1579	.6101	.8752		.0200	.2648	.7339	.6834		.6048	-.3375	-.7797	.4293	1.1743
.0300	-.2931	.5703	.9377		.0500	.1433	.6985	.7387		.7003	-.3375	-.8507	.4089	1.2119
.0400	-.3689	.5484	.9726		.0813	.0146	.6627	.7939						
.0608	-.4351	.5301	1.0022		.1199	-.0181	.6498	.8138						
.0800	-.4593	.5201	1.0184		.1796	-.1136	.6241	.8535						
.1000	-.5271	.5032	1.0463		.2397	-.1798	.6032	.8861						
.1997	-.6007	.4817	1.0825		.2995	-.2492	.5845	.9153						
.2500	-.6168	.4766	1.0911		.3588	-.3284	.5610	.9524						
.2994	-.6326	.4719	1.0992		.4193	-.3841	.5446	.9786						
.3402	-.6231	.4742	1.0953		.4793	-.4072	.5374	.9902						
.3795	-.6288	.4732	1.0970		.5394	-.3091	.5667	.9433						
.4201	-.6418	.4701	1.1023		.5994	-.1300	.6197	.8604						
.4598	-.6811	.4578	1.1235		.6597	.0518	.6723	.7791						
.4996	-.6789	.4589	1.1217		.7203	.2028	.7167	.7102						
.5397	-.7131	.4484	1.1401		.7743	.2843	.7403	.6733						
.5795	-.7569	.4355	1.1631		.8394	.3331	.7545	.6507						
.6197	-.7954	.4257	1.1809		.8996	.3471	.7593	.6430						
.6598	-.8387	.4132	1.2039		.9492	.3089	.7483	.6607						
.6997	-.8731	.4025	1.2239		1.0000	.2113	.7200	.7050						
.7493	-.9234	.5635	.9484											
.8303	-.1513	.6142	.8689											
.8791	-.0506	.6443	.8223											
.9212	.0414	.6703	.7821											
1.0000	.2113	.7200	.7050											

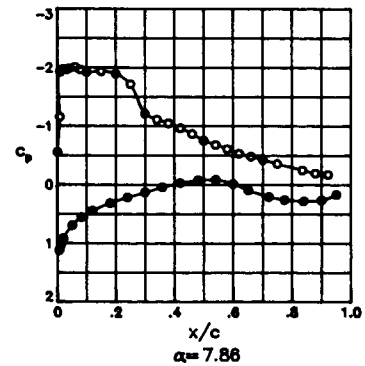
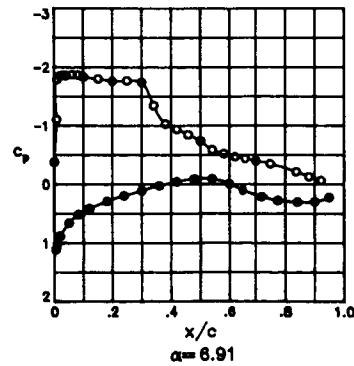
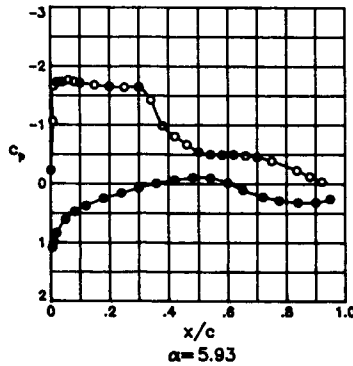
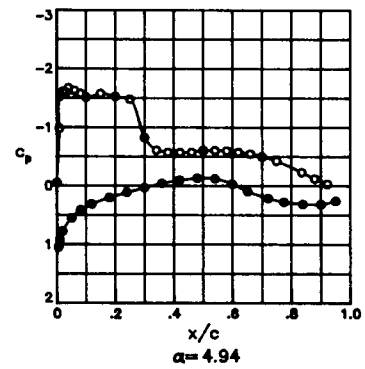
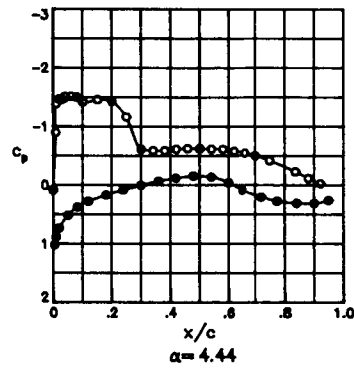
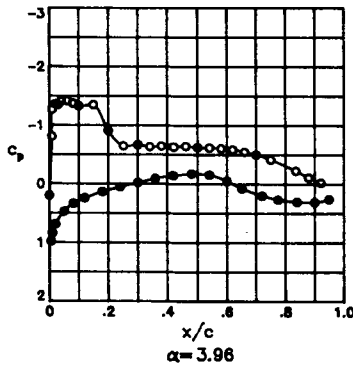
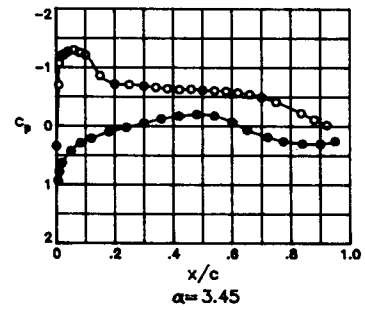
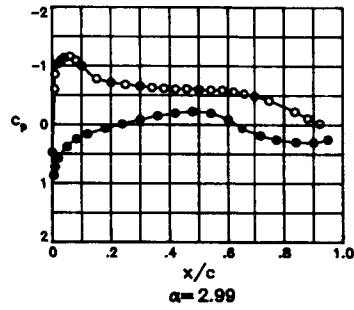
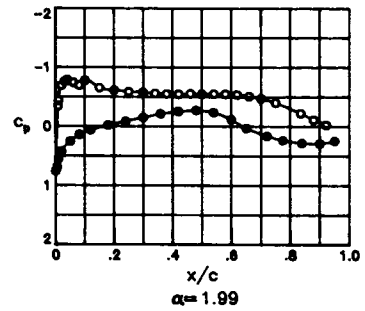
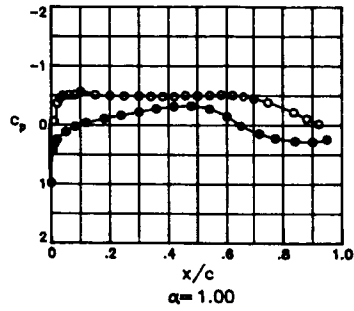
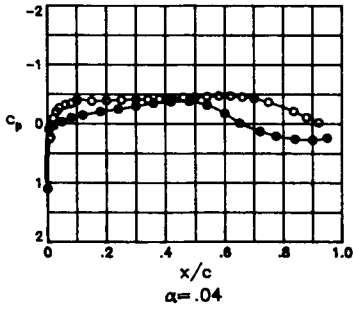
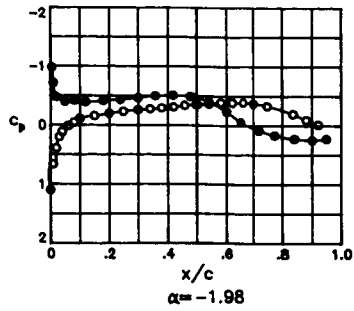
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Appendix H

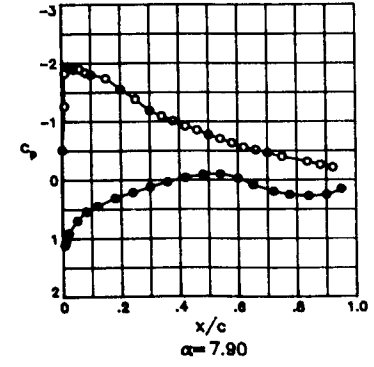
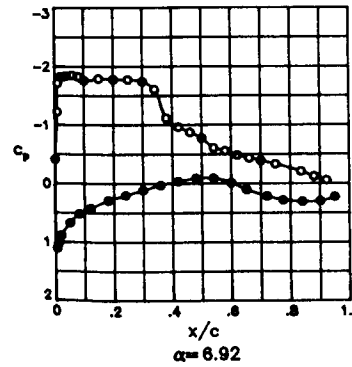
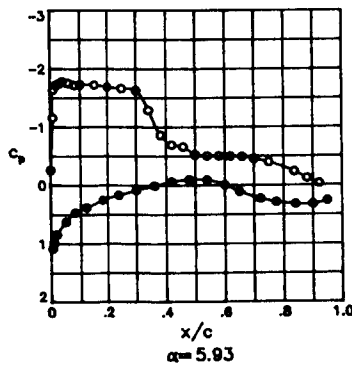
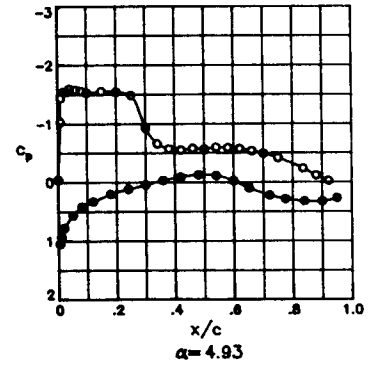
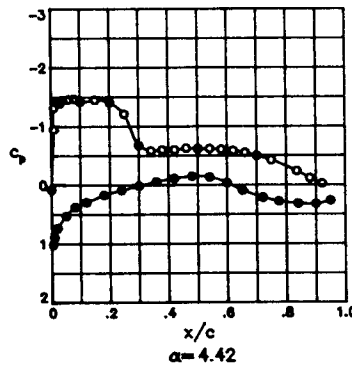
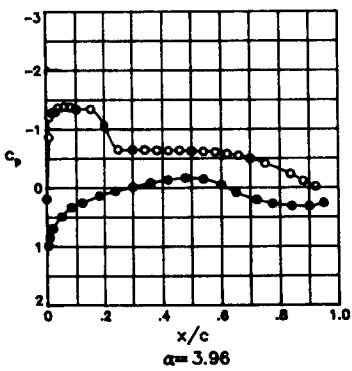
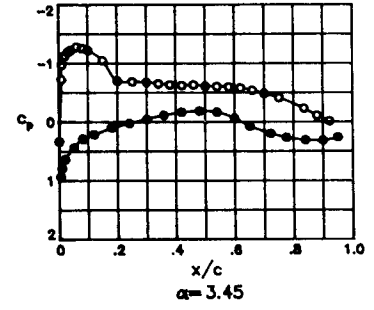
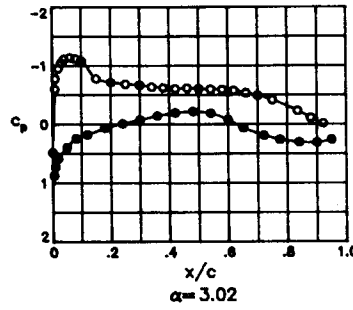
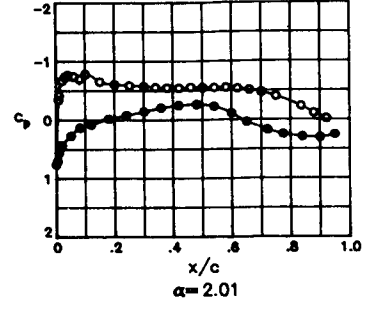
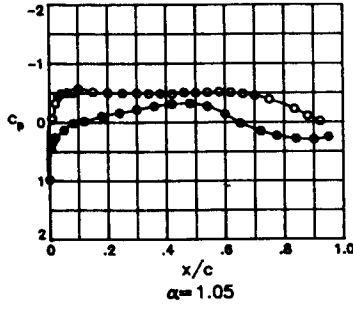
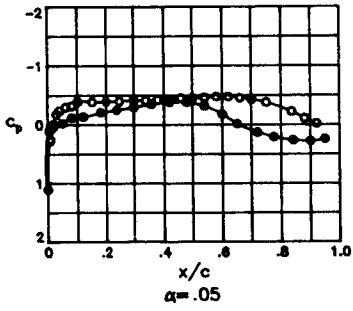
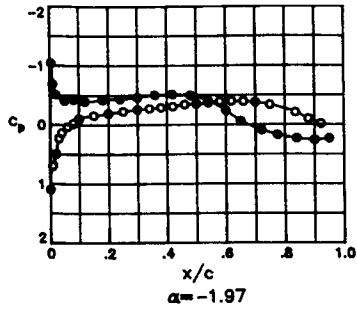
Pressure Data for $M = 0.70$; $R = 4.4 \times 10^6$, 7.7×10^6 , and 14.0×10^6 ; and Fixed Transition

The pressure measurements made on the Boeing BAC I airfoil are presented in coefficient form in graphs and tables in this appendix. The data are for a Mach number of 0.70; Reynolds numbers of 4.4×10^6 , 7.7×10^6 , and 14.0×10^6 ; and fixed transition. The pressure data from the upper surface of the airfoil are plotted as open symbols, and the data from the lower surface are plotted as solid symbols.

TEST 122
 RUN 9
 MACH .704
 R 7.7×10^6



TEST 122
 RUN 13
 MACH .704
 R 14.0×10^6

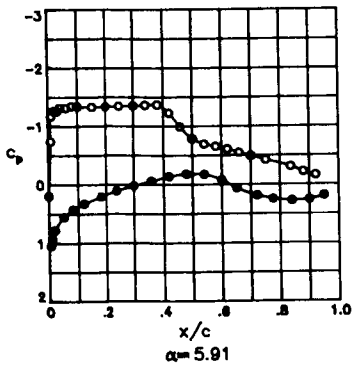
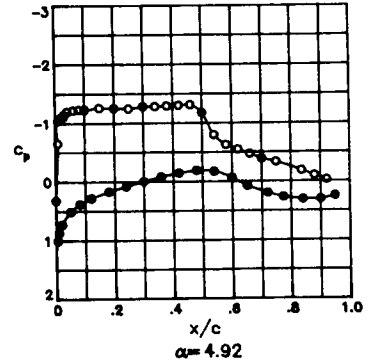
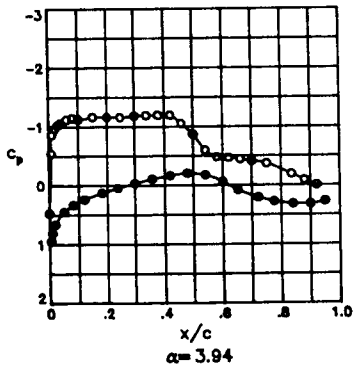
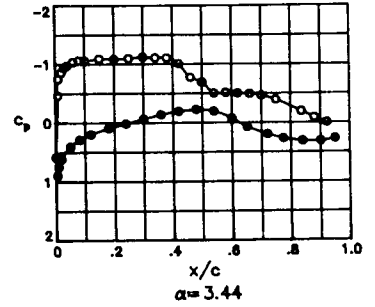
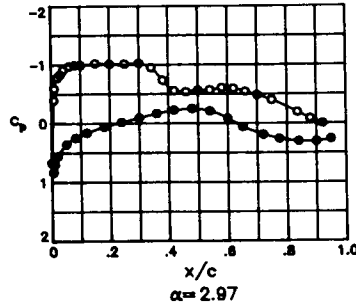
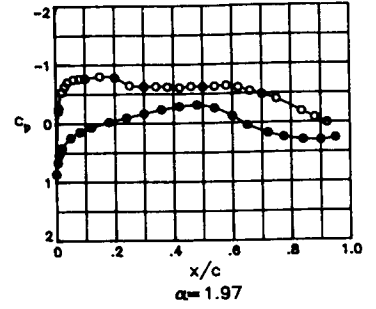
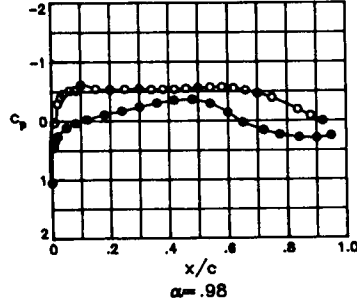
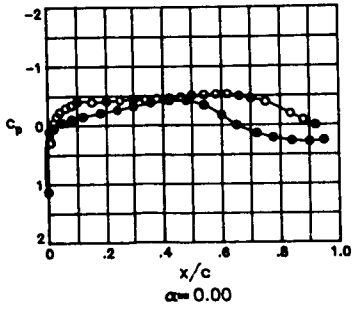
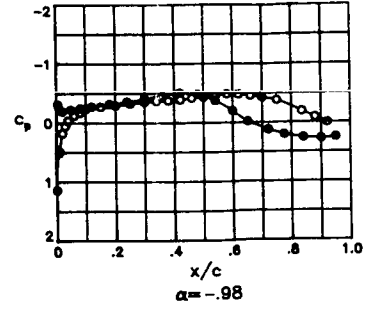
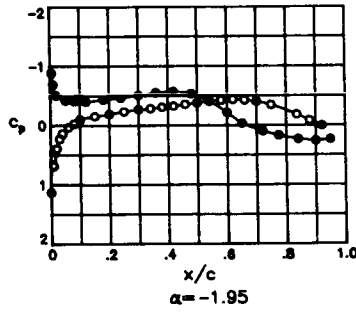


Appendix I

Pressure Data for $M = 0.76$; $R = 4.4 \times 10^6$, 7.7×10^6 , 14.0×10^6 , and 30.0×10^6 ; and Fixed Transition

The pressure measurements made on the Boeing BAC I airfoil are presented in coefficient form in graphs and tables in this appendix. The data are for a Mach number of 0.76; Reynolds numbers of 4.4×10^6 , 7.7×10^6 , 14.0×10^6 , and 30.0×10^6 ; and fixed transition. The pressure data from the upper surface of the airfoil are plotted as open symbols, and the data from the lower surface are plotted as solid symbols.

TEST 122
 RUN 5
 MACH .765
 R 4.4×10^8



TEST 122 PT 17.6049 PSI CN -.3943 C01 .00862 CDCR1 .00850
RUN 5 TT 192.2971 K CM -.0951 C02 .00855 CDCR2 .00842
POINT 4 RC 4.4528 MILLION CC .0009 C03 .00850 CDCR3 .00839
MACH .7575 C04 .01259 CDCR4 .01238
ALPHA .9800 DEG C05 .00832 CDCR5 .00828

UPPER SURFACE LOWER SURFACE
X/C CP P/L/PT MLOC X/C CP P/L/PT MLOC
0.0000 1.0497 .9715 .2034 0.0000 1.0497 .9715 .2034
.0083 .0329 .6910 .7460 .0052 .4318 .8008 .5722
.0097 .0252 .6882 .7499 .0098 .3413 .7779 .6097
.0203 -.2963 .6030 .8615 .0200 .2590 .7555 .6454
.0300 -.3971 .5758 .9241 .0500 .1143 .7152 .7088
.0400 -.4630 .5566 .9543 .0813 .0343 .6944 .7408
.0608 -.5157 .5439 .9747 .1199 -.0277 .6779 .7662
.0800 -.5219 .5428 .9764 .1796 -.1085 .6518 .8063
.1000 -.6110 .5131 1.0247 .2397 -.1700 .6369 .8292
.1498 -.5433 .5343 .9901 .2995 -.2381 .6185 .8575
.1997 -.5345 .5371 .9856 .3518 -.3051 .5983 .8888
.2500 -.5398 .5336 .9912 .4193 -.3514 .5847 .9100
.2994 -.5454 .5312 .9952 .4793 -.3626 .5802 .9172
.3402 -.5347 .5325 .9930 .5334 -.2960 .5999 .8864
.3795 -.5358 .5337 .9911 .5994 -.1415 .6419 .8215
.4201 -.5370 .5326 .9929 .6507 .0253 .6892 .7488
.4598 -.5483 .5311 .9953 .7203 .1586 .7282 .6884
.4996 -.5591 .5315 .9944 .7743 .2370 .7465 .6598
.5397 -.5713 .5229 1.0087 .8394 .2845 .7604 .6377
.5795 -.5753 .5233 1.0080 .8996 .2947 .7657 .6293
.6147 -.5566 .5328 .9926 .9492 .2558 .7547 .6468
.6598 -.5191 .5525 .9766
.6997 -.4691 .5740 .9619
.7493 -.3911 .6286 .9258
.8353 -.1322 .6627 .8367
.8791 -.0045 .6869 .7929
.9212 .0026 .9734 .7556

SPANWISE
X/C Y/R/2 CP P/L/PT MLOC
.0500 -3.375 -3509 .5898 .9021
.3957 -3.375 -5184 .5438 .9748
.5008 -3.375 -5513 .5306 .9962
.6048 -3.375 -5446 .5368 .9862
.7003 -3.375 -4.665 .5604 .9484

TEST 122 PT 17.6410 PSI CN .5305 C01 .00906 CDCR1 .00885
RUN 5 TT 191.9978 K CM -.0955 C02 .00908 CDCR2 .00886
POINT 5 RC 4.4760 MILLION CC -.0067 C03 .00902 CDCR3 .00882
MACH .7617 C04 .01328 CDCR4 .01297
ALPHA 1.9700 DEG C05 .00863 CDCR5 .00856

UPPER SURFACE LOWER SURFACE
X/C CP P/L/PT MLOC X/C CP P/L/PT MLOC
0.0000 .8627 .9205 .3458 0.0000 .8627 .9205 .3458
.0083 -.2155 .6242 .8487 .0052 .6659 .8646 .4604
.0097 -.2674 .6059 .8772 .0098 .5380 .8294 .5237
.0203 -.5402 .5307 .9959 .0200 .4201 .7964 .5795
.0300 -.6310 .5048 1.0383 .0500 .2449 .7479 .6576
.0400 -.7065 .4843 1.0731 .0813 .1447 .7183 .7038
.0608 -.7514 .4686 1.0994 .1199 .0688 .6987 .7342
.0800 -.7623 .4680 1.1006 .1796 -.0223 .6736 .7728
.1498 -.8046 .4439 1.1253 .2397 -.0937 .6518 .8063
.1997 -.7710 .4637 1.1079 .2995 -.1660 .6342 .8334
.2500 -.6441 .4996 1.0470 .3588 -.2360 .6131 .8659
.2994 -.6221 .5108 1.0285 .4193 -.2839 .6040 .8800
.3402 -.6274 .5055 1.0373 .4793 -.3084 .5940 .8955
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.4201 -.6056 .5154 1.0210 .5994 -.1149 .6506 .8081
.4598 -.6235 .5082 1.0327 .6507 .0426 .6926 .7436
.4996 -.6135 .5125 1.0256 .7203 .1747 .7301 .6856
.5397 -.6729 .5077 1.0334 .7743 .2488 .7492 .6555
.5795 -.6034 .5630 1.0414 .8394 .2950 .7617 .6365
.6147 -.6034 .5171 1.0180 .8996 .3013 .7659 .6289
.6598 -.5379 .5499 .9966 .9492 .2652 .7531 .6493
.6997 -.4421 .5694 .9717
.7493 -.4014 .6297 .9319
.8353 -.1408 .6547 .8404
.8791 -.0020 .6810 .7963
.9212 .0044 .9155 .7587

SPANWISE
X/C Y/R/2 CP P/L/PT MLOC
.0500 -3.375 -5570 .5305 .9962
.3957 -3.375 -6077 .5135 1.0241
.5008 -3.375 -6079 .5137 1.0237
.6048 -3.375 -6078 .5102 1.0295
.7003 -3.375 -4.476 .5454 .9722

TEST 122 PT 17.6574 PSI CN .6731 C01 .01092 CDCR1 .01048
RUN 5 TT 192.3425 K CM -.0916 C02 .01110 CDCR2 .01061
POINT 6 RC 4.4405 MILLION CC -.0170 C03 .01080 CDCR3 .01043
MACH .7592 C04 .01544 CDCR4 .01506
ALPHA 2.9694 DEG C05 .00957 CDCR5 .00924

UPPER SURFACE LOWER SURFACE
X/C CP P/L/PT MLOC X/C CP P/L/PT MLOC
0.0000 .6687 .8654 .4580 0.0000 .6687 .8659 .4580
.0083 -.3489 .5707 .9321 .0052 .8324 .9115 .3681
.0097 -.5444 .5176 1.0172 .0098 .6965 .8750 .4408
.0203 -.7795 .4695 1.0979 .0200 .5561 .8349 .5142
.0300 -.8341 .4504 1.1311 .0500 .3573 .7806 .6054
.0400 -.9091 .4313 1.1654 .0813 .2450 .7501 .6540
.0608 -.9744 .4144 1.1983 .1199 .1588 .7266 .6910
.0800 -.9910 .4101 1.2042 .1796 .0589 .6994 .7331
.1498 -.9944 .4094 1.2048 .2397 -.0199 .6764 .7685
.1997 -1.203 .4003 1.2226 .2995 -.0952 .6562 .7994
.2500 -1.0123 .4036 1.2165 .3588 -.1687 .6358 .8309
.2994 -1.0149 .4026 1.2182 .4193 -.2234 .6212 .8534
.3402 -1.0745 .4006 1.2222 .4793 -.2505 .6148 .8633
.3795 -.9622 .4220 1.1822 .5394 -.2112 .6245 .8482
.4201 -.7343 .4405 1.0791 .5994 -.0972 .6607 .7926
.4598 -.5383 .5355 .9811 .6507 .0643 .7007 .7310
.4996 -.5621 .5288 .9900 .7203 .1913 .7359 .6764
.5397 -.5707 .5275 1.0092 .7743 .2637 .7536 .6486
.5795 -.6056 .5153 1.0210 .8394 .3071 .7669 .6273
.6147 -.5927 .5221 1.0094 .8996 .3084 .7690 .6239
.6598 -.5402 .5490 .9924 .9492 .2641 .7549 .6467
.6997 -.4426 .5734 .9666
.7493 -.3980 .6297 .9288
.8353 -.1434 .6609 .8390
.8791 -.0046 .6939 .7925
.9212 -.0002 .8698 .7564

SPANWISE
X/C Y/R/2 CP P/L/PT MLOC
.0500 -3.375 -7110 .4859 1.0700
.3957 -3.375 -7912 .4621 1.1108
.5008 -3.375 -5635 .5278 1.0007
.6048 -3.375 -5861 .5186 1.0137
.7003 -3.375 -4.4831 .5497 .9654

TEST	132	PT	17.6529	PSI	CM	.9769	CD1	.06557	CDCDR1	.06422
RUN	5	TT	192.3968	K	CM	-1.094	CD2	.06814	CDCDR2	.06698
POINT	11	RC	4.4472	MILLION	CC	-.0281	CD3	.06171	CDCDR3	.06055
		MACH	.7614				CD4	.08015	CDCDR4	.07894
		ALPHA	5.9093	DEG			CD5	.05852	CDCDR5	.03700

UPPER SURFACE					LOWER SURFACE					SPANWISE				
X/C	CP	P _L /PT	MLOC		X/C	CP	P _L /PT	MLOC		X/C	Y/S/2	CP	P _L /PT	MLOC
0.0000	.1906	.7339	.6796		0.0000	.1906	.7339	.6796		.0500	-.3375	-.9545	.4139	1.1971
.0033	-.7424	.4758	1.0872		.0052	1.0476	.9708	.2062		.3957	-.3375	-1.2787	.3260	1.3739
.0077	-1.1724	.3571	1.3078		.0098	.9268	.9381	.3034		.5068	-.3375	-.7667	.4680	1.1004
.0203	-1.2664	.3351	1.3541		.0200	.7775	.8964	.3982		.6048	-.3375	-.5982	.5148	1.0218
.0300	-1.2641	.3330	1.3585		.0500	.5549	.8345	.5148		.7003	-.3375	-.4802	.5445	.9737
.0400	-1.3199	.3161	1.3959		.0813	.4261	.7989	.5753						
.0608	-1.3165	.3170	1.3938		.1199	.3257	.7727	.6180						
.0800	-1.3493	.3132	1.4024		.1796	.2052	.7391	.6714						
.1000	-1.3410	.3132	1.4424		.2397	.1695	.7101	.7165						
.1498	-1.3311	.3105	1.4085		.2995	.0199	.6863	.7533						
.1997	-1.3358	.3111	1.4672		.3588	-.0593	.6673	.7826						
.2500	-1.3472	.3134	1.4420		.4193	-.1378	.6444	.8177						
.2994	-1.3530	.3094	1.4111		.4793	-.1794	.6335	.8344						
.3402	-1.3636	.3076	1.4151		.5394	-.1716	.6356	.8311						
.3795	-1.3716	.3054	1.4204		.5994	-.0769	.6598	.7940						
.4201	-1.2299	.3409	1.3416		.6507	.0642	.6999	.7324						
.4598	-.9406	.4490	1.2062		.7203	.1885	.7348	.6780						
.4996	-.7476	.4663	1.1635		.7743	.2490	.7489	.6560						
.5397	-.6442	.4872	1.0678		.8394	.2786	.7554	.6457						
.5795	-.6537	.4964	1.0523		.8996	.2579	.7521	.6509						
.6197	-.6484	.5124	1.0259		.9492	.1901	.7331	.6808						
.6598	-.5477	.5420	.9991											
.6997	-.4907	.5622	.9782											
.7473	-.4207	.5931	.9452											
.8353	-.3062	.6199	.8971											
.8791	-.2481	.6384	.8547											
.9212	-.1584	.7344	.8265											

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TEST 122 PT 17.5701 PSI CN .9376
 RUN 8 TT 129.0370 K CM -1.089
 POINT 11 RC 7.8360 MILLION CC -.0284
 MACH .7617
 ALPHA 4.9100 DEG

CD1 .03416 CDCOR1 .03305
 CD2 .03430 CDCOR2 .03300
 CD3 .03333 CDCOR3 .03195
 CD4 .04523 CDCOR4 .04416
 CD5 .02443 CDCOR5 .02390

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC
0.0000	.3431	.7753	.6143	0.0000	.3431	.7753	.6143
.0043	-.6278	.3063	1.0364	.0052	.9978	.9566	.2524
.0197	-.9855	.4071	1.2103	.0098	.8701	.9229	.3406
.0203	-1.1362	.3729	1.2763	.0200	.7171	.8860	.4315
.0300	-1.1298	.3713	1.2801	.0500	.5078	.8223	.5363
.0400	-1.1920	.3539	1.3149	.0813	.3713	.7836	.6007
.0608	-1.2097	.3465	1.3303	.1199	.2753	.7584	.6414
.0800	-1.2422	.3404	1.3432	.1796	.1646	.7285	.6884
.1000	-1.2291	.3453	1.3328	.2397	.0691	.7002	.7323
.1498	-1.2362	.3394	1.3453	.2995	-.0094	.6802	.7631
.1997	-1.2504	.3386	1.3469	.3588	-.0943	.6557	.8008
.2500	-1.2741	.3375	1.3494	.4193	-.1594	.6394	.8259
.2994	-1.2714	.3336	1.3577	.4793	-.2089	.6243	.8491
.3402	-1.2726	.3308	1.3637	.5394	-.1886	.6277	.8438
.3795	-1.2744	.3267	1.3728	.5994	-.0736	.6591	.7956
.4201	-1.2873	.3222	1.3827	.6507	.0703	.7013	.7306
.4598	-1.3154	.3190	1.3897	.7203	.1962	.7348	.6786
.4996	-1.2625	.3309	1.3636	.7743	.2617	.7501	.6544
.5397	-1.0450	.3852	1.2521	.8394	.3063	.7643	.6320
.5795	-.6918	.4673	1.0686	.8996	.3028	.7639	.6325
.6197	-.5490	.5278	1.0011	.9492	.2457	.7483	.6573
.6598	-.4380	.5751	.9733				
.6997	-.3957	.5852	.9257				
.7493	-.3354	.6244	.9497				
.8353	-.1458	.6514	.8493				
.8791	-.1343	.6668	.8672				
.9212	-.0360	.7755	.7831				

SPANWISE				
X/C	Y/8/2	CP	P _L /PT	MLOC
.0500	-.3375	-1.0567	.3868	1.2490
.3957	-.3375	-1.2088	.3465	1.3303
.5008	-.3375	-1.2151	.3445	1.3345
.6048	-.3375	-.5422	.5323	.9938
.7003	-.3375	-.4051	.5678	.9371

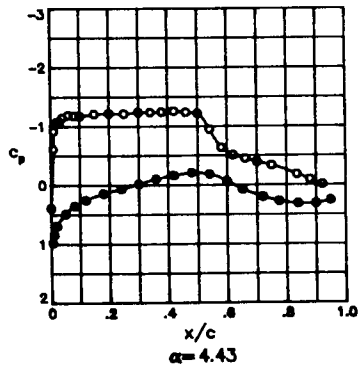
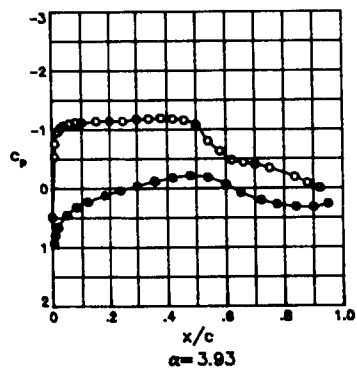
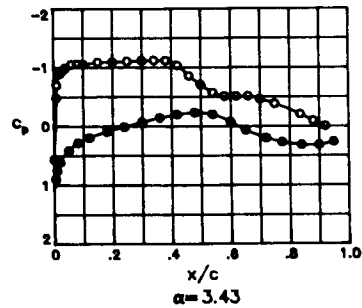
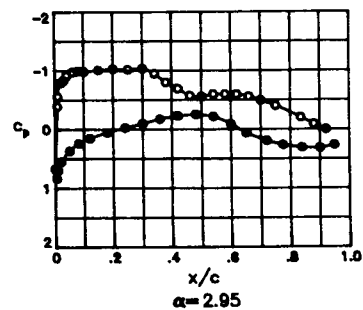
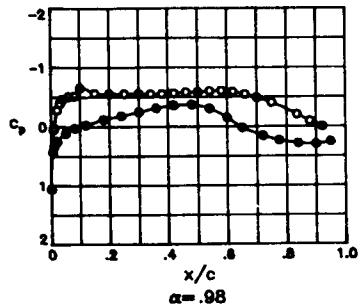
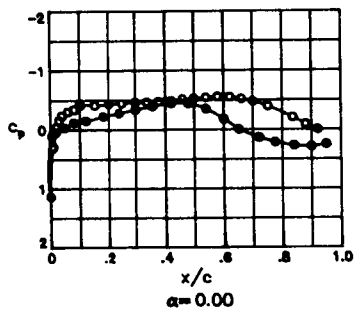
TEST 122 PT 17.5755 PSI CN .9640
 RUN 8 TT 130.4408 K CM -1.151
 POINT 12 RC 7.7324 MILLION CC -.0240
 MACH .7649
 ALPHA 5.9107 DEG

CD1 .06555 CDCOR1 .06391
 CD2 .06799 CDCOR2 .06681
 CD3 .06067 CDCOR3 .05882
 CD4 .07909 CDCOR4 .07772
 CD5 .03839 CDCOR5 .03726

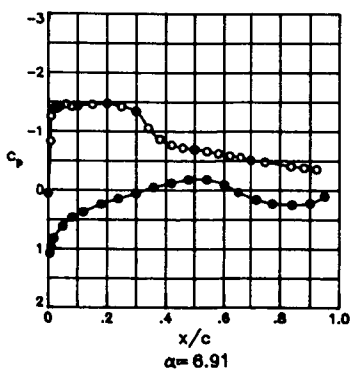
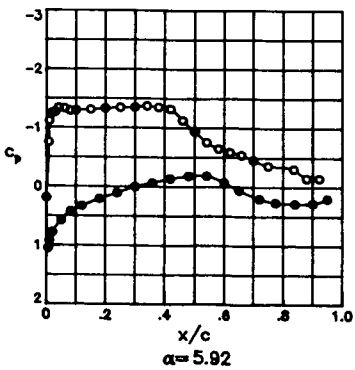
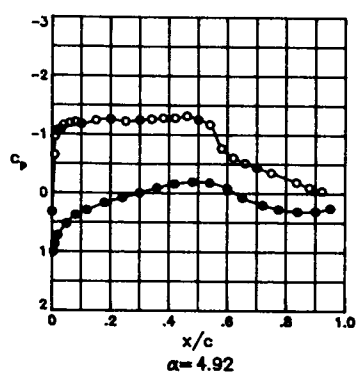
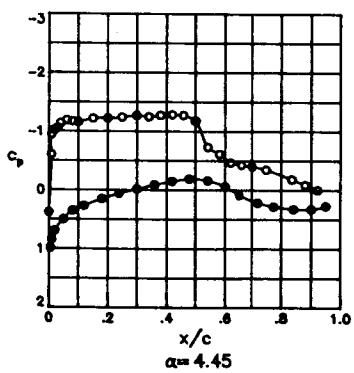
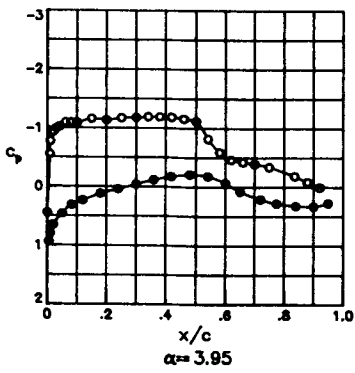
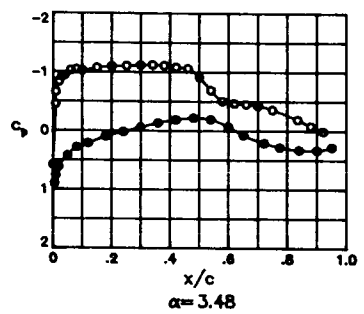
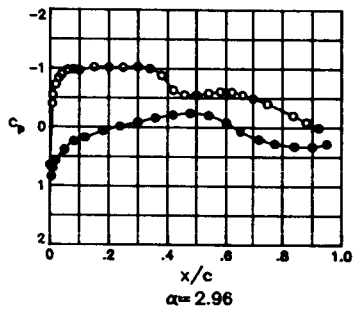
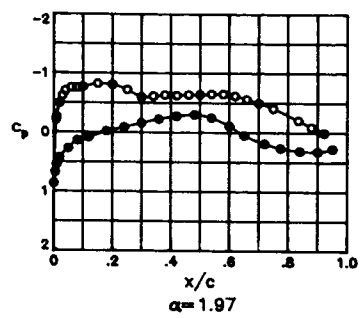
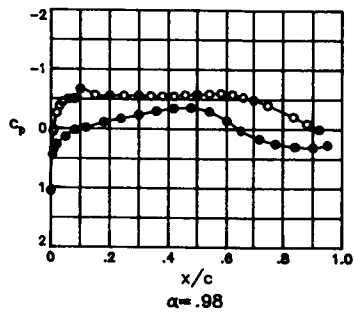
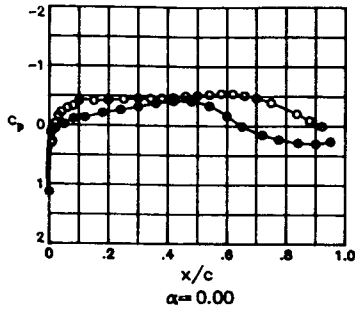
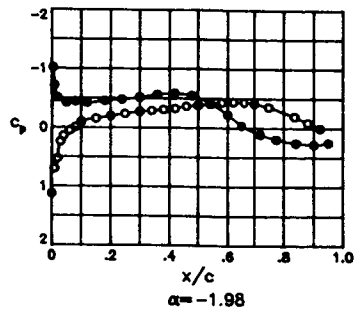
UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC
0.0000	.2148	.7399	.6705	0.0000	.2148	.7399	.6705
.0083	-.7092	.4841	1.0736	.0052	1.0410	.9685	.2144
.0097	-1.1345	.3647	1.2928	.0098	.9133	.9328	.3168
.0203	-1.2147	.3413	1.3412	.0200	.7694	.8930	.4054
.0300	-1.2293	.3380	1.3482	.0500	.5572	.8353	.5137
.0400	-1.3181	.3174	1.3935	.0813	.4123	.7939	.5839
.0608	-1.3036	.3175	1.3931	.1199	.3113	.7657	.6297
.0800	-1.3060	.3165	1.3954	.1796	.1930	.7334	.6807
.1000	-1.3007	.3193	1.3890	.2397	.1014	.7090	.7187
.1498	-1.3153	.3172	1.3939	.2995	.0127	.6845	.7565
.1997	-1.3221	.3134	1.3979	.3588	-.0753	.6580	.7972
.2500	-1.3112	.3144	1.4001	.4193	-.1599	.6323	.8367
.2994	-1.3151	.3096	1.4109	.4793	-.2119	.6190	.8573
.3402	-1.3243	.3090	1.4124	.5394	-.1945	.6260	.8465
.3795	-1.3334	.3084	1.4137	.5994	-.0851	.6571	.7986
.4201	-1.2856	.3248	1.3768	.6507	.0475	.6924	.7444
.4598	-.9164	.4245	1.1780	.7203	.1701	.7243	.6949
.4996	-.7149	.4769	1.0857	.7743	.2458	.7481	.6576
.5397	-.7269	.4785	1.0831	.8394	.2694	.7526	.6505
.5795	-.6548	.4947	1.0556	.8996	.2594	.7517	.6520
.6197	-.6217	.5072	1.0348	.9492	.1654	.7242	.6951
.6598	-.5800	.5496	1.0103				
.6997	-.4847	.5596	.9666				
.7493	-.4191	.5920	.9494				
.8353	-.3057	.6066	.8994				
.8791	-.2530	.5957	.8762				
.9212	-.2544	.7399	.8930				

SPANWISE				
X/C	Y/8/2	CP	P _L /PT	MLOC
.0500	-.3375	-1.1327	.3626	1.2970
.3957	-.3375	-1.2769	.3206	1.3862
.5008	-.3375	-.7795	.4594	1.1159
.6048	-.3375	-.6044	.5129	1.0255
.7003	-.3375	-.4646	.5511	.9636

TEST 122
RUN 10
MACH .765
R 7.7×10^6



TEST 122
 RUN 12
 MACH .765
 R 14.0×10^6



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TEST 122 PT 21.9357 PSI CN .9805
RUN 12 TT 101.4644 K CM -.1095
POINT 10 KC 13.9090 MILLION CC -.0292
MACH .7603
ALPHA 5.9200 DEG

CD1 .05847 CDCOR1 .05716
CD2 .05437 CDCOR2 .06280
CD3 .07262 CDCOR3 .07109
CD4 .08104 CDCOR4 .07988
CD5 .04097 CDCOR5 .04003

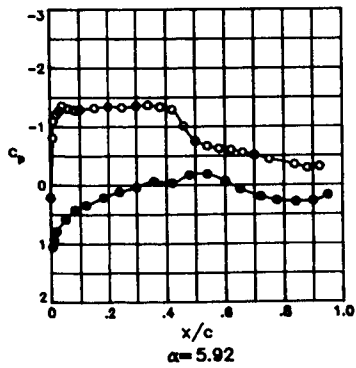
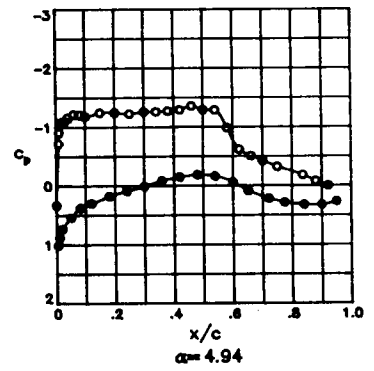
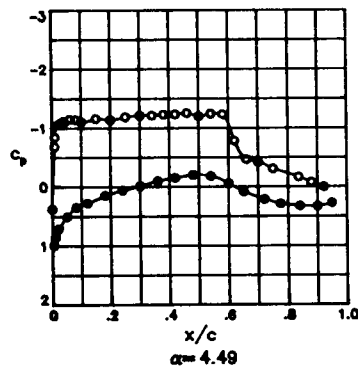
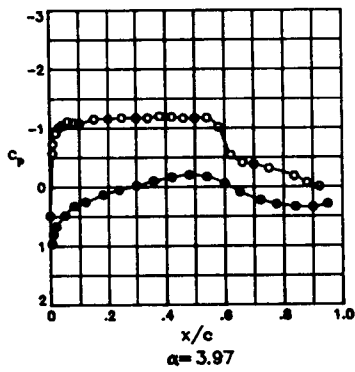
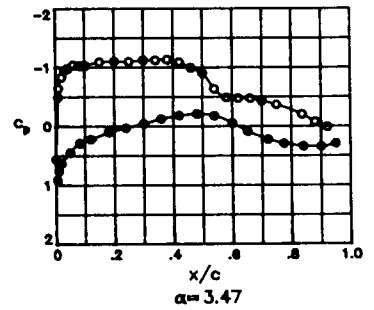
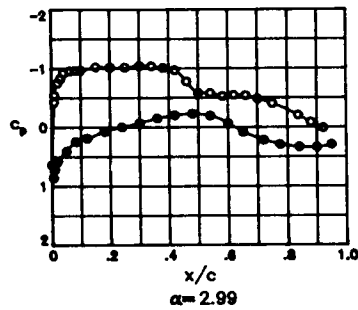
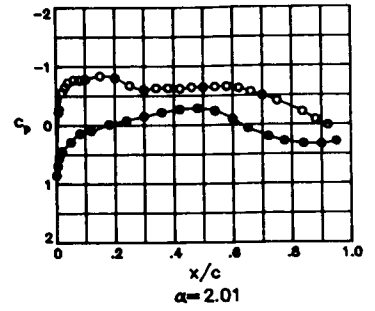
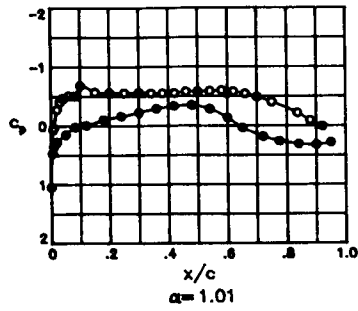
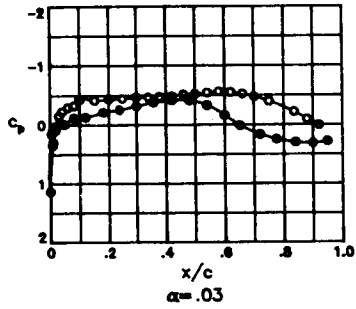
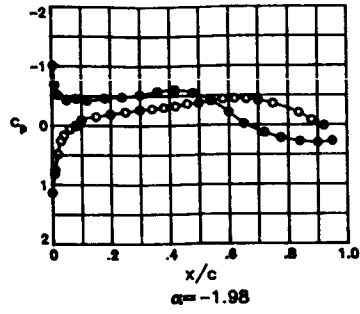
UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _s L/PT	MLOC	X/C	CP	P _s L/PT	MLOC	X/C	Y/B/2	CP	P _s L/PT	MLOC
.0000	.1884	.7342	.6804	.0000	.1884	.7342	.6804	.0500	-.3375	-1.0809	.3887	1.2464
.0083	-.7519	.4751	1.0899	.0052	1.0489	.9714	.2045	.3957	-.3375	-1.3133	.3212	1.3858
.0097	-1.1176	.3752	1.2727	.0098	.9230	.9369	.3071	.5008	-.3375	-1.0207	.4067	1.2119
.0203	-1.2502	.3379	1.3494	.0200	.7783	.8973	.3971	.6048	-.3375	-.5948	.5175	1.0190
.0300	-1.2676	.3357	1.3542	.0500	.5761	.8424	.5019	.7003	-.3375	-.4388	.5649	.9427
.0400	-1.3366	.3186	1.3916	.0813	.4233	.8010	.5730					
.0608	-1.3271	.3224	1.3831	.1199	.3329	.7735	.6180					
.0800	-1.2892	.3258	1.3757	.1796	.2080	.7414	.6690					
.1000	-1.2947	.3297	1.3671	.2397	.1083	.7103	.7176					
.1498	-1.3074	.3182	1.3923	.2995	.0018	.6823	.7608					
.1997	-1.3291	.3152	1.3991	.3598	-.0652	.6658	.7861					
.2500	-1.3428	.3152	1.3992	.4193	-.1381	.6440	.8196					
.2994	-1.3483	.3104	1.4100	.4793	-.1807	.6363	.8315					
.3402	-1.3710	.3112	1.4082	.5394	-.1828	.6318	.8385					
.3795	-1.3484	.3105	1.4097	.5994	-.0677	.6652	.7871					
.4201	-1.3222	.3208	1.3866	.6507	.0579	.6975	.7374					
.4598	-1.1283	.3700	1.2831	.7203	.1987	.7388	.6732					
.4996	-.9371	.4275	1.1736	.7743	.2662	.7591	.6410					
.5397	-.7549	.4809	1.0800	.8394	.2840	.7598	.6400					
.5795	-.6444	.5034	1.0422	.8996	.2764	.7577	.6432					
.6197	-.5856	.5197	1.0154	.9492	.1953	.7361	.6773					
.6598	-.5394	.5581	.9925									
.6997	-.4569	.5835	.9531									
.7493	-.3590	.5971	.9128									
.8353	-.3046	.6426	.8425									
.8791	-.1487	.6417	.8208									
.9212	-.1495	.7345	.8238									

TEST 122 PT 21.9363 PSI CN .9734
RUN 12 TT 100.9572 K CM -.1079
POINT 11 RC 13.9990 MILLION CC -.0248
MACH .7586
ALPHA 6.9100 DEG

CD1 .07433 CDCOR1 .07384
CD2 .07315 CDCOR2 .07253
CD3 .04398 CDCOR3 .04348
CD4 .05145 CDCOR4 .05088
CD5 .04618 CDCOR5 .04603

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P _s L/PT	MLOC	X/C	CP	P _s L/PT	MLOC	X/C	Y/B/2	CP	P _s L/PT	MLOC
.0000	.0572	.6598	.7338	.0000	.0572	.6598	.7338	.0500	-.3375	-1.1263	.3745	1.2742
.0083	-.8297	.4567	1.1217	.0052	1.0813	.9807	.1673	.3957	-.3375	-.8143	.4596	1.1166
.0097	-1.2553	.3408	1.3432	.0098	.9670	.9493	.2741	.5008	-.3375	-.7091	.4902	1.0642
.0203	-1.3892	.3034	1.4262	.0200	.8249	.9103	.3694	.6048	-.3375	-.6040	.5127	1.0268
.0300	-1.3836	.3049	1.4228	.0500	.6180	.8538	.4813	.7003	-.3375	-.5285	.5389	.9842
.0400	-1.4297	.2931	1.4505	.0813	.4657	.8144	.5504					
.0608	-1.4602	.2927	1.4537	.1199	.3700	.7873	.5955					
.0800	-1.4252	.2975	1.4399	.1796	.2377	.7530	.6507					
.1000	-1.4338	.3000	1.4341	.2397	.1446	.7280	.6900					
.1498	-1.4570	.2942	1.4477	.2995	.0536	.7054	.7251					
.1997	-1.4756	.2937	1.4490	.3598	-.0487	.6723	.7761					
.2500	-1.4204	.2976	1.4397	.4193	-.1208	.6531	.8056					
.2994	-1.3428	.3198	1.3889	.4793	-.1815	.6368	.8308					
.3402	-1.0557	.3984	1.2276	.5394	-.1825	.6366	.8310					
.3795	-.8641	.4508	1.1320	.5994	-.0934	.6583	.7976					
.4201	-.7690	.4730	1.0933	.6507	.0375	.6933	.7438					
.4598	-.7213	.4846	1.0736	.7203	.1626	.7284	.6894					
.4996	-.6949	.4931	1.0593	.7743	.2295	.7475	.6595					
.5397	-.6620	.5034	1.0422	.8394	.2477	.7503	.6551					
.5795	-.6221	.5104	1.0366	.8996	.2229	.7435	.6657					
.6197	-.5848	.5209	1.0134	.9492	.1037	.7131	.7132					
.6598	-.5578	.5451	.9954									
.6997	-.5091	.5544	.9740									
.7493	-.4768	.5740	.9594									
.8353	-.4050	.5823	.9282									
.8791	-.3774	.5847	.9152									
.9212	-.3542	.7003	.9117									

TEST 122
 RUN 14
 MACH .765
 R 30.0×10^6



TEST 122	PT	44.2513	PSI	CM	1.0064	CD1	.06631	CDCOR1	.06521
RUN 16	TT	97.3708	K	CM	-.1259	CD2	.07228	CDCOR2	.07106
POINT 2	RC	29.7970	MILLION	CC	-.0240	CD3	.06959	CDCOR3	.06829
	MACH	5.7586				CD4	.09154	CDCOR4	.09012
	ALPHA	5.9200	DEG			CD5	.04598	CDCOR5	.04541

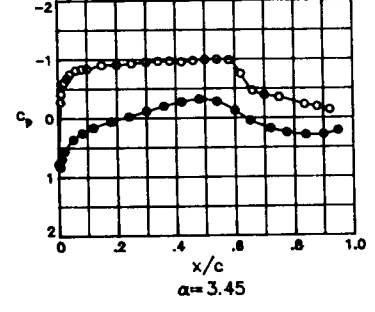
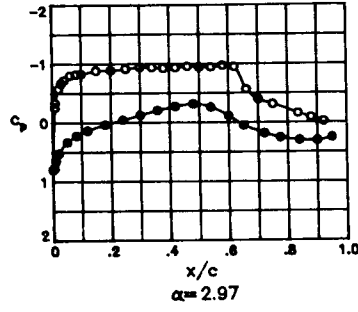
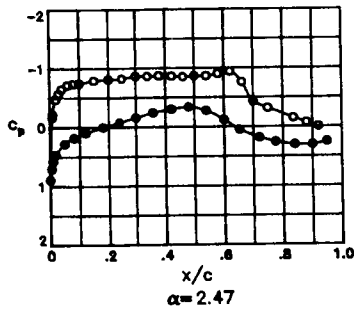
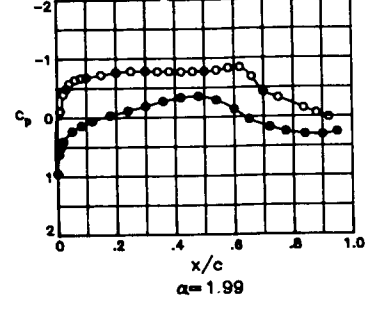
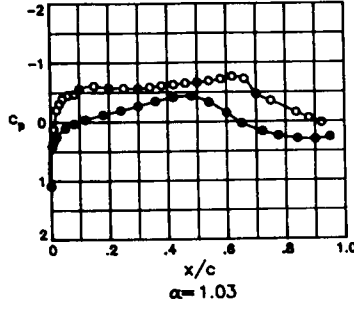
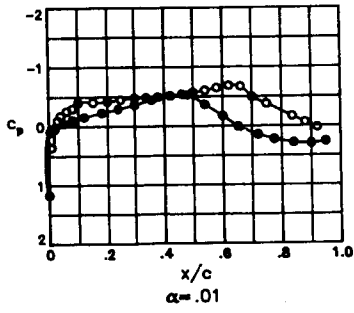
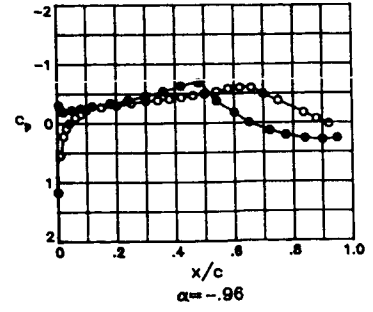
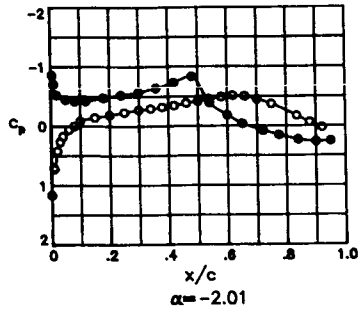
UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/8/2	CP	P/L/PT	MLOC
0.0000	.2141	.7397	.6732	0.0000	.2141	.7397	.6732	.0500	-.3375	-.8847	.4437	1.1467
.0003	-.8104	.4564	1.1242	.0052	1.0578	.9745	.1933	.3957	-.3375	-1.3465	.3189	1.3927
.0097	-1.0976	.3872	1.2514	.0098	.9329	.9399	.2999	.5008	-.3375	-.9401	.4316	1.1684
.0203	-1.2105	.3536	1.3183	.0200	.7873	.9005	.3013	.6048	-.3375	-.5997	.5183	1.0197
.0306	-1.2739	.3380	1.3511	.0500	.5850	.8461	.4965	.7003	-.3375	-.5014	.5460	.9748
.0400	-1.3080	.3176	1.3956	.0813	.4183	.7995	.5768					
.0698	-1.3124	.3266	1.3758	.1199	.3368	.7756	.6160					
.0800	-1.2924	.3279	1.3730	.1746	.2067	.7410	.6712					
.1000	-1.2476	.3291	1.3704	.2397	.1154	.7143	.7129					
.1498	-1.3229	.3187	1.3932	.2995	.0316	.6937	.7448					
.1997	-1.3449	.3175	1.3958	.3548	-.0669	.6653	.7886					
.2500	-1.3345	.3176	1.3956	.4193	-.0373	.6736	.7758					
.2994	-1.3556	.3121	1.4080	.4793	-.1748	.6392	.8289					
.3402	-1.3692	.3142	1.4034	.5394	-.1888	.6308	.8418					
.3745	-1.3404	.3141	1.4035	.5994	-.0702	.6666	.7867					
.4201	-1.2461	.3320	1.3639	.6577	.0672	.7052	.7271					
.4598	-1.0133	.4113	1.2055	.7203	.1948	.7398	.6730					
.4996	-.7554	.4813	1.0813	.7743	.2617	.7565	.6467					
.5397	-.6700	.5417	1.0470	.8394	.2859	.7623	.6373					
.5795	-.6254	.5125	1.0292	.8996	.2722	.7591	.6425					
.6197	-.6027	.5196	1.0174	.9492	.1757	.7325	.6846					
.6598	-.5588	.5430	.9984									
.6997	-.5125	.5436	.9786									
.7493	-.4431	.5871	.9472									
.8353	-.3493	.6049	.9094									
.8791	-.2895	.5490	.8624									
.9212	-.3134	.7405	.8906									

Appendix J

Pressure Data for $M = 0.80$; $R = 4.4 \times 10^6$, 7.7×10^6 , and 14.0×10^6 ; and Fixed Transition

The pressure measurements made on the Boeing BAC I airfoil are presented in coefficient form in graphs and tables in this appendix. The data are for a Mach number of 0.80; Reynolds numbers of 4.4×10^6 , 7.7×10^6 , and 14.0×10^6 ; and fixed transition. The pressure data from the upper surface of the airfoil are plotted as open symbols, and the data from the lower surface are plotted as solid symbols.

TEST 122
 RUN 3
 MACH .807
 R 4.4×10^6



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TEST 122 PT 17.6481 PSI CN .6919
 RUN 4 TT 196.2893 K CM -.1132
 POINT 7 RC 4.4324 MILLION CC -.0085
 MACH .7977
 ALPHA 2.9663 DEG

CD1 .01709 CDCOR1 .01641
 CD2 .01768 CDCOR2 .01639
 CD3 .01832 CDCOR3 .01732
 CD4 .02725 CDCOR4 .02667
 CD5 .01825 CDCOR5 .01728

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC
0.0000	.7978	.8917	.4177	0.0000	.7978	.8917	.4077
.0083	-.2604	.8626	.9133	.0052	.7718	.8828	.4236
.0097	-.3360	.8570	.9537	.0098	.6463	.8473	.4920
.0203	-.5703	.4918	1.0601	.0200	.5234	.8123	.5529
.0300	-.6686	.4653	1.1657	.0500	.3281	.7530	.6495
.0400	-.7231	.4446	1.1415	.0813	.2226	.7238	.6954
.0608	-.8020	.4245	1.1776	.1199	.1352	.6967	.7373
.0800	-.8185	.4171	1.1913	.1796	.0334	.6675	.7821
.1000	-.8288	.4151	1.1949	.2397	-.0469	.6440	.8183
.1498	-.8422	.3994	1.2243	.2995	-.1301	.6196	.8558
.1997	-.8907	.3969	1.2291	.3588	-.2088	.5970	.8910
.2500	-.9126	.3911	1.2403	.4193	-.2750	.5794	.9183
.2994	-.9445	.3843	1.2536	.4793	-.3210	.5629	.9444
.3402	-.9381	.3819	1.2581	.5394	-.2648	.5789	.9191
.3795	-.9243	.3853	1.2514	.5994	-.1157	.6231	.8504
.4201	-.9346	.3830	1.2559	.6507	.0471	.6711	.7766
.4598	-.9584	.3764	1.2688	.7203	.1807	.7098	.7170
.4996	-.9428	.3802	1.2614	.7743	.2546	.7317	.6829
.5397	-.9473	.3793	1.2631	.8394	.2977	.7428	.6656
.5795	-.9721	.3688	1.2840	.8996	.3022	.7456	.6611
.6197	-.9487	.3788	1.2641	.9492	.2493	.7301	.6655
.6598	-.9586	.3740	1.0612				
.6997	-.9358	.3661	.9724				
.7493	-.9159	.6114	.9393				
.8353	-.9109	.6291	.8668				
.8791	-.0910	.6489	.8416				
.9212	-.0228	.8957	.8694				

SPANWISE				
X/C	Y/B/2	CP	P _L /PT	MLOC
.0500	-.3375	-.5565	.4940	1.0563
.3957	-.3375	-.8957	.3910	1.2403
.5008	-.3375	-.9551	.3793	1.2631
.6048	-.3375	-1.0067	.3599	1.3021
.7003	-.3375	-.3975	.5413	.9789

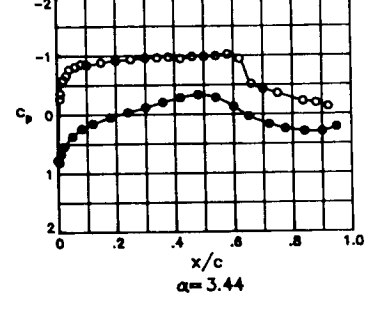
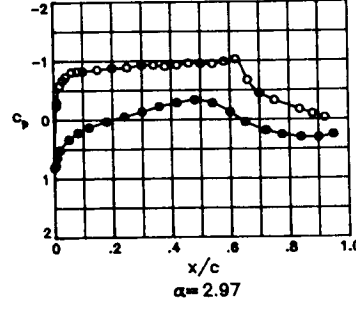
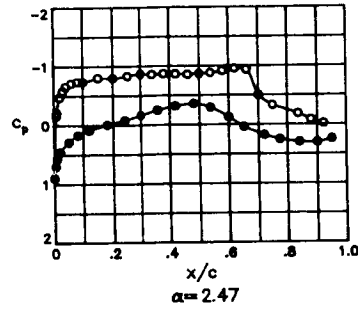
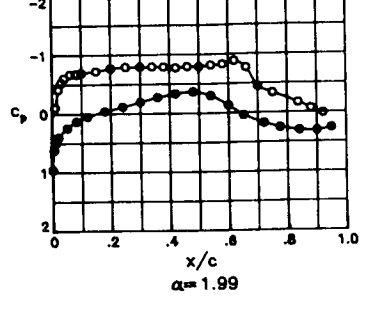
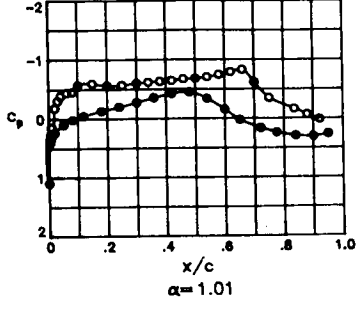
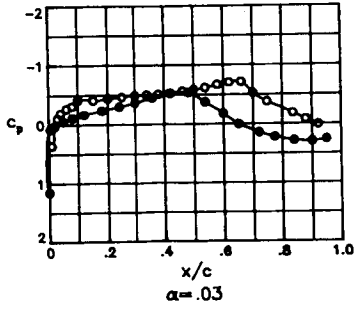
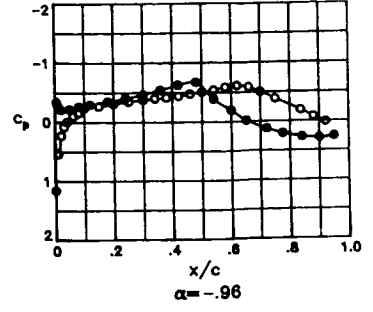
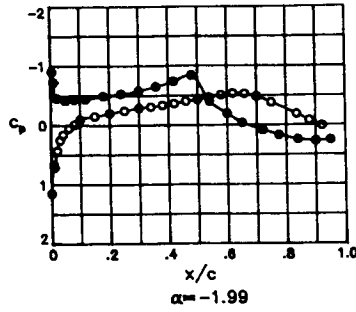
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 RUN 4 TT 195.6515 K CM -.1195
 POINT 8 PC 4.4843 MILLION CC -.0048
 MACH .8072
 ALPHA 3.4499 DEG

CD1 .02670 CDCOR1 .02582
 CD2 .02834 CDCOR2 .02746
 CD3 .03171 CDCOR3 .03089
 CD4 .04599 CDCOR4 .04521
 CD5 .02947 CDCOR5 .02814

UPPER SURFACE				LOWER SURFACE			
X/C	CP	P _L /PT	MLOC	X/C	CP	P _L /PT	MLOC
0.0000	.7759	.8826	.4260	0.0000	.7759	.8826	.4260
.0083	-.2780	.8705	.9223	.0052	.8237	.8979	.3951
.0097	-.4139	.8519	.9940	.0098	.6901	.8579	.4730
.0203	-.6357	.4753	1.0880	.0200	.5571	.8182	.5430
.0300	-.6769	.4532	1.1262	.0500	.3620	.7603	.6380
.0400	-.7487	.4316	1.1646	.0813	.2505	.7270	.6904
.0608	-.8157	.4113	1.2020	.1199	.1599	.6989	.7338
.0800	-.8327	.4040	1.2156	.1796	.0550	.6671	.7827
.1000	-.8395	.4010	1.2213	.2397	-.0298	.6447	.8171
.1498	-.9032	.3861	1.2499	.2995	-.1191	.6170	.8598
.1997	-.9077	.3833	1.2553	.3588	-.2067	.5916	.8992
.2500	-.9288	.3775	1.2661	.4193	-.2787	.5701	.9331
.2994	-.9545	.3698	1.2819	.4793	-.3215	.5569	.9538
.3402	-.9696	.3648	1.2921	.5394	-.2838	.5672	.9376
.3795	-.9726	.3626	1.2965	.5994	-.1289	.6139	.8647
.4201	-.9587	.3678	1.2861	.6507	.0406	.6638	.7879
.4598	-.9770	.3617	1.2984	.7203	.1741	.7040	.7260
.4996	-.9918	.3585	1.3049	.7743	.2462	.7261	.6918
.5397	-.9991	.3577	1.3066	.8394	.2839	.7341	.6792
.5795	-.9855	.3554	1.3114	.8996	.2782	.7331	.6808
.6197	-.9532	.4259	1.1750	.9492	.2136	.7150	.7089
.6598	-.8666	.5331	1.0244				
.6997	-.8912	.5498	.9917				
.7493	-.8538	.5827	.9655				
.8353	-.8245	.5928	.9135				
.8791	-.1961	.6080	.8971				
.9212	-.1427	.8629	.8733				

SPANWISE				
X/C	Y/B/2	CP	P _L /PT	MLOC
.0500	-.3375	-.5406	.4909	1.0616
.3957	-.3375	-.9227	.3772	1.2674
.5008	-.3375	-.9674	.3670	1.2877
.6048	-.3375	-.8700	.3900	1.2424
.7003	-.3375	-.4053	.5302	.9968

TEST 122
 RUN 7
 MACH .807
 R 7.7×10^6



TEST 122 PT 17.9954 PSI CN .6940
 RUN 7 TT 132.5392 K CM -.1184
 POINT 7 PC 7.7566 MILLION CC -.0067
 MACH .7997
 ALPHA 2.9677 DEG

CD1 .01926 CDCOR1 .01829
 CD2 .02020 CDCOR2 .01924
 CD3 .02028 CDCOR3 .01931
 CD4 .02999 CDCOR4 .02919
 CD5 .01989 CDCOR5 .01912

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/2	CP	P/L/PT	MLOC
0.0000	.8118	.8950	.4613	0.0000	.8118	.8950	.4613	.0500	-.3375	-.6424	.4701	1.0974
.6083	-.2387	.5664	.9070	.0052	.7641	.8804	.4305	.3957	-.3375	-.9076	.3906	1.2421
.0097	-.2974	.5682	.9364	.0098	.6409	.8462	.4944	.5008	-.3375	-.9550	.3748	1.2726
.0203	-.5837	.4891	1.0651	.0200	.5137	.8090	.5588	.6048	-.3375	-1.0167	.3569	1.3086
.0300	-.6708	.4634	1.1089	.0500	.3344	.7552	.6464	.7003	-.3375	-.4468	.9227	1.0094
.0400	-.7324	.4426	1.1455	.0813	.2172	.7232	.6967					
.0608	-.8115	.4236	1.1797	.1199	.1298	.6959	.7389					
.0800	-.8245	.4167	1.1923	.1796	.0300	.6685	.7810					
.1000	-.8296	.4182	1.1897	.2397	-.0576	.6386	.8271					
.1498	-.8524	.4047	1.2148	.2995	-.1381	.6156	.8624					
.1997	-.8781	.3982	1.2270	.3588	-.2273	.5875	.9062					
.2500	-.8881	.3926	1.2378	.4193	-.2901	.5702	.9333					
.2994	-.9291	.3822	1.2579	.4793	-.3354	.5559	.9559					
.3402	-.9276	.3814	1.2594	.5394	-.2847	.5688	.9355					
.3795	-.9063	.3849	1.2527	.5994	-.1307	.6158	.8621					
.4201	-.9266	.3810	1.2603	.6507	.0402	.6669	.7834					
.4598	-.9585	.3728	1.2764	.7203	.1839	.7126	.7131					
.4996	-.9525	.3808	1.2606	.7743	.2545	.7296	.6866					
.5397	-.9486	.3749	1.2724	.8394	.2988	.7423	.6688					
.5795	-.9863	.3629	1.2963	.8996	.3046	.7471	.6592					
.6197	-1.0252	.3581	1.3062	.9492	.2552	.7323	.6823					
.6598	-.6727	.5246	1.1136									
.6997	-.4497	.5628	1.0065									
.7493	-.3277	.6067	.9441									
.8353	-.1751	.6274	.8775									
.8791	-.0994	.6471	.8445									
.9212	-.0316	.8956	.8135									

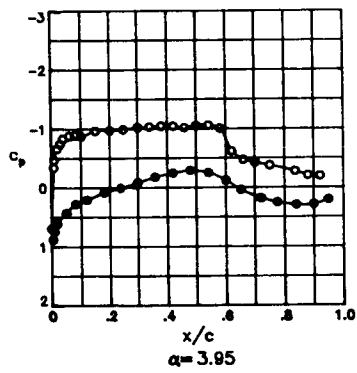
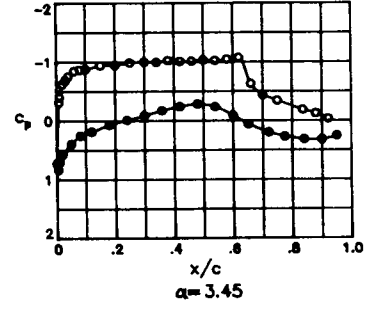
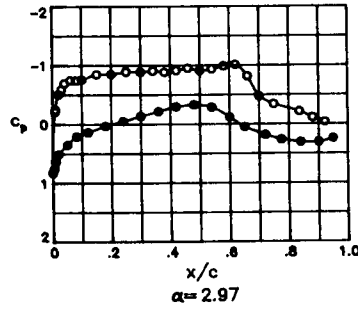
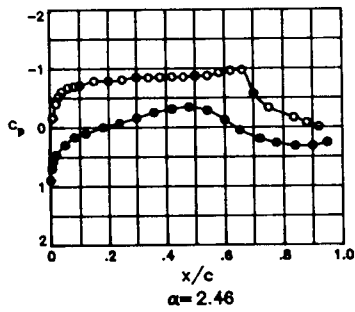
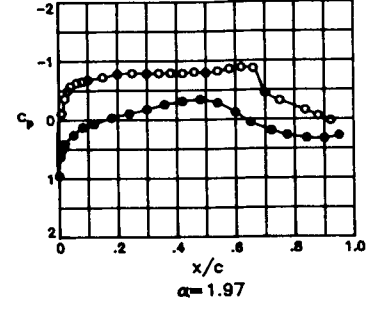
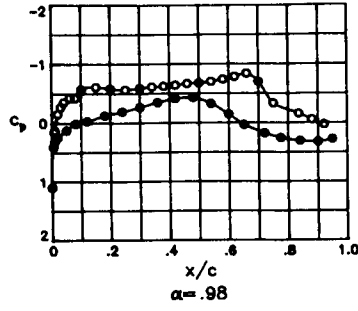
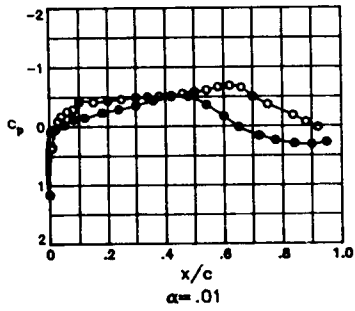
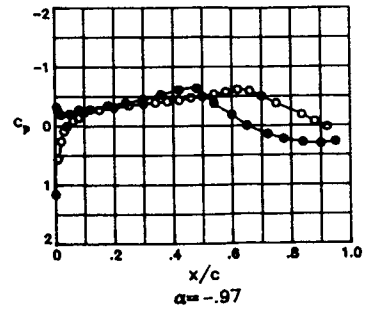
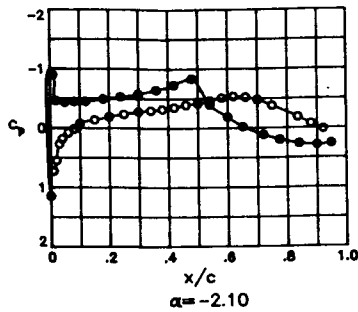
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TEST 122 PT 17.9946 PSI CN .7278
 RUN 7 TT 132.7910 K CM -.1241
 POINT 8 RC 7.7723 MILLION CC -.0046
 MACH .8058
 ALPHA 3.4380 DEG

CD1 .02724 CDCOR1 .02605
 CD2 .02849 CDCOR2 .02696
 CD3 .03197 CDCOR3 .03112
 CD4 .04581 CDCOR4 .04490
 CD5 .02958 CDCOR5 .02721

UPPER SURFACE				LOWER SURFACE				SPANWISE				
X/C	CP	P/L/PT	MLOC	X/C	CP	P/L/PT	MLOC	X/C	Y/B/2	CP	P/L/PT	MLOC
0.0000	.7715	.8816	.4283	0.0000	.7715	.8816	.4283	.0500	-.3375	-.6690	.4547	1.1240
.6083	-.2734	.5727	.9294	.0052	.8105	.4934	.4046	.3957	-.3375	-.9095	.3751	1.2719
.0097	-.3646	.5466	.9707	.0098	.6711	.4515	.4848	.5008	-.3375	-.9776	.3649	1.2923
.0203	-.5949	.4768	1.0859	.0200	.5420	.4127	.5526	.6048	-.3375	-1.0189	.3489	1.3253
.0300	-.6653	.4345	1.1244	.0500	.3690	.7643	.6318	.7003	-.3375	-.4429	.5244	1.0066
.0400	-.7091	.4297	1.1666	.0813	.2372	.7233	.6955					
.0608	-.8157	.4118	1.2015	.1199	.1539	.7002	.7323					
.0800	-.8606	.4013	1.2212	.1796	.0497	.6699	.7769					
.1000	-.8493	.4052	1.2138	.2397	-.0375	.6418	.8222					
.1498	-.8861	.3905	1.2418	.2995	-.1207	.6194	.8566					
.1997	-.9197	.3840	1.2544	.3588	-.2077	.5927	.8980					
.2500	-.9381	.3770	1.2681	.4193	-.2862	.5700	.9336					
.2994	-.9630	.3703	1.2814	.4793	-.3367	.5526	.9612					
.3402	-.9677	.3656	1.2909	.5394	-.2873	.5670	.9384					
.3795	-.9787	.3620	1.2982	.5994	-.1353	.6105	.8704					
.4201	-.9518	.3676	1.2869	.6507	.0325	.6631	.7893					
.4598	-.9869	.3613	1.2986	.7203	.1687	.7005	.7318					
.4996	-.9856	.3568	1.3089	.7743	.2449	.7256	.6929					
.5397	-.9962	.3585	1.3054	.8394	.2927	.7410	.6688					
.5795	-1.0218	.3536	1.3155	.8996	.2912	.7408	.6691					
.6197	-.9460	.3764	1.2894	.9492	.2196	.7166	.7070					
.6598	-.5169	.5271	1.0506									
.6997	-.4307	.5449	1.0020									
.7493	-.3584	.5882	.9729									
.8353	-.2246	.5921	.9058									
.8791	-.1977	.6134	.8988									
.9212	-.1343	.8811	.8664									

TEST 122
 RUN 11
 MACH .807
 R 14.0×10^6



TEST 122 PT 20.5076 PSI CM .6499
RUN 11 TT 99.4960 K CM -1241
POINT 7 TC 13.8560 MILLION CC -.0037
MACH .8064
ALPHA 2.9665 DEG

UPPER SURFACE LOWER SURFACE SPANWISE
X/C CP P/L/PT MLOC X/C CP P/L/PT MLOC X/C Y/B/2 CP P/L/PT MLOC
0.0000 .8246 .8963 .3950 0.0000 .8246 .8983 .3950 .0500 -.3375 -.5939 .4745 1.0908
.0083 -.2195 .5918 .9005 .0052 .7547 .8786 .4405 .3957 -.3375 -.8962 .3887 1.2463
.0097 -.2536 .5766 .9243 .0098 .6366 .8422 .5022 .5008 -.3375 -.9423 .3720 1.2790
.0203 -.5098 .5643 1.0407 .0200 .5101 .8036 .5686 .6048 -.3375 -1.0008 .3558 1.3118
.0300 -.5936 .4768 1.0869 .0500 .3484 .7586 .6419 .7003 -.3375 -.4511 .5167 1.0203
.0400 -.6964 .4518 1.1361 .0813 .2064 .7158 .7090 .1199 .1367 .6910 .7474
.0608 -.7439 .4289 1.1710 .1796 .0286 .6609 .7937 .2397 -.0526 .6379 .8290
.1000 -.7606 .4271 1.1743 .2995 -.1395 .6103 .8716 .3588 -.2183 .5914 .9010
.1498 -.8467 .4033 1.2184 .4193 -.2093 .5605 .9496 .6098 .7037 .8649 .5328
.1997 -.8526 .3988 1.2359 .4793 -.3324 .5534 .9609 .5394 -.2917 .5638 .9444
.2500 -.8896 .3941 1.2399 .5394 -.2917 .5638 .9444 .5994 -.1225 .6146 .8650
.2994 -.8402 .3844 1.2546 .5994 -.1225 .6146 .8650 .6537 .0464 .6666 .7848
.3402 -.9060 .3833 1.2567 .6537 .0464 .6666 .7848 .7203 .1807 .7029 .7290
.3795 -.8844 .3875 1.2486 .7203 .1807 .7029 .7290 .7743 .2597 .7278 .6903
.4201 -.9118 .3800 1.2632 .7743 .2597 .7278 .6903 .8394 .3053 .7428 .6668
.4598 -.9514 .3715 1.2801 .8394 .3053 .7428 .6668 .8996 .3049 .7414 .6690
.4996 -.9097 .3775 1.2682 .8996 .3049 .7414 .6690 .9492 .2423 .7255 .6939
.5397 -.9375 .3717 1.2796 .9492 .2423 .7255 .6939 .5795 -.9874 .3600 1.3032
.5795 -.9874 .3600 1.3032 .6197 -.16121 .3464 1.3241
.6197 -.16121 .3464 1.3241 .6598 -.8158 .5160 1.1996
.6598 -.8158 .5160 1.1996 .6997 -.4748 .5512 1.0215
.7493 -.3413 .5838 .9641
.8353 -.2181 .6203 .9132
.8791 -.1149 .6434 .8558
.9212 -.0354 .8988 .8205

TEST 122 PT 20.5075 PSI CM .7729
RUN 11 TT 99.5817 K CM -1268
POINT 8 TC 13.7340 MILLION CC -.0090
MACH .7963
ALPHA 3.4500 DEG

UPPER SURFACE LOWER SURFACE SPANWISE
X/C CP P/L/PT MLOC X/C CP P/L/PT MLOC X/C Y/B/2 CP P/L/PT MLOC
0.0000 .7208 .8898 .4516 0.0000 .7208 .8698 .4516 .0500 -.3375 -.7039 .4572 1.1207
.0083 -.3152 .5684 .9372 .0052 .8318 .9012 .3891 .3957 -.3375 -.9766 .3756 1.2719
.0097 -.4455 .5392 .9837 .0200 .5704 .8248 .5328 .5008 -.3375 -.9988 .3658 1.2915
.0203 -.6229 .4793 1.0831 .0500 .3908 .7716 .6210 .6048 -.3375 -1.0894 .3404 1.3440
.0300 -.6981 .4568 1.1215 .0813 .2528 .7340 .6806 .7003 -.3375 -.4424 .5319 .9955
.0400 -.7565 .4355 1.1592 .1199 .1838 .7136 .7125 .1498 -.9405 .3877 1.2483
.0608 -.8478 .4141 1.1982 .1796 .0686 .6824 .7607 .2397 -.0129 .6571 .7996
.0800 -.8680 .4075 1.2105 .2995 -.0934 .6318 .8384 .3588 -.1758 .6120 .8691
.1000 -.8792 .4081 1.2093 .4193 -.2435 .5903 .9027 .6098 .7037 .8649 .5328
.1498 -.9405 .3877 1.2483 .4793 -.3324 .5534 .9609 .5394 -.2917 .5638 .9444
.1997 -.9477 .3774 1.2684 .5394 -.2917 .5638 .9444 .5994 -.1225 .6146 .8650
.2500 -.9874 .3774 1.2684 .6537 .0464 .6666 .7848 .7203 .1807 .7029 .7290
.2994 -.9941 .3710 1.2811 .7203 .1807 .7029 .7290 .7743 .2597 .7278 .6903
.3402 -.9968 .3672 1.2887 .7743 .2597 .7278 .6903 .8394 .3053 .7428 .6668
.3795 -.10243 .3659 1.2913 .8394 .3053 .7428 .6668 .8996 .3049 .7414 .6690
.4201 -.10687 .3717 1.2797 .8996 .3049 .7414 .6690 .9492 .2518 .7335 .6815
.4598 -.10153 .3646 1.2938 .9492 .2518 .7335 .6815 .5795 -.9874 .3600 1.3032
.4996 -.10312 .3595 1.3042 .6197 -.16121 .3464 1.3241
.5397 -.10226 .3632 1.2968 .6598 -.8158 .5160 1.1996
.5795 -.10439 .3555 1.3126 .6997 -.4387 .5586 1.0619
.6197 -.10741 .3441 1.3274 .7493 -.3455 .6151 .9527
.6598 -.8297 .5278 1.0865 .8353 -.1898 .6223 .8797
.6997 -.4387 .5586 1.0619 .8791 -.1290 .6492 .8538
.7493 -.3455 .6151 .9527 .9212 -.0352 .8700 .8112
.8353 -.1898 .6223 .8797
.8791 -.1290 .6492 .8538
.9212 -.0352 .8700 .8112

TEST 122 PT 20.5053 PSI CM .7823
RUN 11 TT 99.5484 K CM -1270
POINT 4 TC 13.7940 MILLION CC -.0061
MACH .8020
ALPHA 3.9497 DEG

UPPER SURFACE LOWER SURFACE SPANWISE
X/C CP P/L/PT MLOC X/C CP P/L/PT MLOC X/C Y/B/2 CP P/L/PT MLOC
0.0000 .8864 .8584 .4730 0.0000 .8864 .8584 .4730 .0500 -.3375 -.7300 .4463 1.1400
.0083 -.3462 .5560 .9568 .0052 .8781 .9145 .3602 .3957 -.3375 -.9921 .3622 1.2987
.0097 -.4786 .5174 1.0192 .0098 .7387 .8739 .4437 .5008 -.3375 -1.0366 .3552 1.3132
.0203 -.6669 .4629 1.1109 .0200 .6060 .8354 .5143 .6048 -.3375 -.7005 .4497 1.1340
.0300 -.7499 .4393 1.1524 .0500 .4291 .7855 .5985 .7003 -.3375 -.4256 .5309 .9972
.0400 -.8365 .4180 1.1911 .0813 .2772 .7386 .6734 .1199 .2054 .7175 .7064
.0608 -.8821 .3992 1.2261 .1796 .0808 .6809 .7631 .2397 .0031 .6589 .7967
.0800 -.8958 .3950 1.2341 .2995 -.0849 .6311 .8395 .3588 -.1806 .6036 .8820
.1000 -.8797 .3993 1.2259 .4193 -.2490 .5848 .9114 .6098 .7037 .8649 .5328
.1498 -.9686 .3747 1.2736 .4793 -.3324 .5534 .9609 .5394 -.2917 .5638 .9444
.1997 -.9753 .3696 1.2839 .5394 -.2917 .5638 .9444 .5994 -.1225 .6146 .8650
.2500 -.9903 .3661 1.2968 .6537 .0464 .6666 .7848 .7203 .1807 .7029 .7290
.2994 -.1.0239 .3581 1.3072 .7203 .1807 .7029 .7290 .7743 .2597 .7278 .6903
.3402 -1.0324 .3541 1.3155 .7743 .2597 .7278 .6903 .8394 .3053 .7428 .6668
.3795 -1.0475 .3510 1.3218 .8394 .3053 .7428 .6668 .8996 .3049 .7414 .6690
.4201 -1.0443 .3526 1.3185 .8996 .3049 .7414 .6690 .9492 .2033 .7143 .6744
.4598 -1.0210 .3511 1.3216 .9492 .2033 .7143 .6744 .5795 -.9874 .3600 1.3032
.4996 -1.0525 .3452 1.3338 .6197 -.16121 .3464 1.3241
.5397 -1.0618 .3411 1.3426 .6598 -.8158 .5160 1.1996
.5795 -1.0082 .3553 1.3128 .6997 -.4180 .4707 1.0974
.6197 -.6180 .4707 1.0974 .7493 -.4764 .5290 1.0249
.6598 -.4764 .5290 1.0249 .8353 -.2796 .5965 .9276
.6997 -.4257 .5421 1.0000 .8791 -.2099 .5947 .8035
.7493 -.3777 .5743 .9792 .9212 -.1478 .8589 .9960
.8353 -.2796 .5965 .9276
.8791 -.2099 .5947 .8035
.9212 -.1478 .8589 .9960

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TABLE I. AIRFOIL COORDINATES

Airfoil coordinates		
x, percent chord	y, percent chord	
	Upper surface	Lower surface
0.00	0.000	0.000
.05	.242	-.235
.12	.388	-.349
.20	.504	-.432
.30	.619	-.508
.50	.798	-.625
.80	1.006	-.758
1.20	1.229	-.903
1.80	1.502	-1.084
2.40	1.731	-1.240
3.20	1.993	-1.424
4.00	2.222	-1.590
5.00	2.474	-1.779
6.00	2.697	-1.954
7.00	2.896	-2.116
8.00	3.079	-2.269
10.00	3.401	-2.548
12.00	3.678	-2.799
14.00	3.921	-3.028
16.00	4.134	-3.236
19.00	4.412	-3.519
22.00	4.647	-3.767
26.00	4.907	-4.048
30.00	5.113	-4.269
35.00	5.308	-4.463
40.00	5.436	-4.548
45.00	5.499	-4.501
50.00	5.498	-4.285
55.00	5.424	-3.875
60.00	5.269	-3.265
65.00	5.013	-2.508
70.00	4.638	-1.704
74.00	4.245	-1.107
77.00	3.896	-.709
80.00	3.501	-.379
83.00	3.060	-.123
85.00	2.742	.005
87.00	2.407	.100
89.00	2.060	.162
91.00	1.705	.190
93.00	1.350	.185
95.00	.994	.147
97.00	.639	.075
98.00	.461	.026
99.00	.283	-.030
100.00	.106	-.096

TABLE II. PRESSURE TAP COORDINATES

Orifice	x/c	y/(b/2)
Upper surface		
1	0.000	0.176
2	.008	-.263
3	.110	-.138
4	.020	-.160
5	.030	-.188
6	.040	-.213
7	.061	-.238
8	.080	-.113
9	.100	
10	.150	
11	.200	
12	.250	
13	.299	
14	.340	
15	.379	
16	.420	
17	.460	
18	.500	
19	.540	
20	.580	
21	.620	
22	.660	
23	.700	
24	.749	-.263
25	.799	-.240
26	.835	-.213
27	.879	-.188
28	.921	-.163
29	1.000	-.138
Lower surface		
1	0.005	0.159
2	.010	.147
3	.020	.130
4	.050	.116
5	.081	.098
6	.120	.075
7	.180	
8	.240	
9	.300	
10	.359	
11	.419	
12	.479	
13	.539	
14	.599	
15	.651	
16	.720	
17	.774	.157
18	.839	.115
19	.900	.068
20	.949	.028
Additional spanwise orifices		
1	0.050	-0.338
2	.395	
3	.501	
4	.605	
5	.700	

TABLE III. BAC I TEST RESULTS

[There are no runs 1, 2, 42, and 54]

(a) Fixed transition

Point	M	R	α	c_n	c_m	c_d
Run 3						
1	0.795	4.317×10^6	-2.05	-0.041	-0.089	0.01077
2	.802	4.466	-.98	.124	-.097	.00933
3	.799	4.460	.02	.268	-.101	^a .00918
Run 4						
4	0.796	4.438×10^6	1.03	0.411	-0.103	^a 0.00943
5	.802	4.474	1.05	.415	-.103	^a .00982
6	.801	4.450	2.03	.560	-.107	.01178
7	.798	4.432	3.02	.692	-.113	^a .01646
8	.807	4.484	3.51	.718	-.119	^a .02763
10	.808	4.528	2.50	.634	-.117	^a .01853
^b 11	.802	4.453	2.52	.632	-.114	^a .01522
Run 5						
1	0.758	4.456×10^6	-1.99	-0.010	-0.088	0.00855
2	.758	4.458	-1.00	.126	-.092	.00821
3	.758	4.459	.00	.261	-.095	.00823
4	.757	4.453	1.00	.394	-.095	.00842
5	.762	4.476	2.01	.530	-.096	.00886
6	.759	4.440	3.02	.673	-.092	^a .01075
7	.760	4.431	3.50	.759	-.095	^a .01306
8	.758	4.435	4.01	.833	-.098	^a .01766
10	.760	4.442	5.01	.930	-.104	^a .03518
11	.761	4.447	6.02	.977	-.109	^a .06703
Run 6						
1	0.696	4.459×10^6	-2.05	-0.015	-0.084	0.00841
2	.696	4.462	.04	.245	-.089	.00805
3	.698	4.484	1.04	.367	-.090	.00814
4	.698	4.481	2.05	.493	-.090	.00842
5	.697	4.481	3.05	.616	-.088	.00876
6	.697	4.461	3.52	.675	-.085	.00942
7	.697	4.452	4.03	.751	-.083	.01089
8	.698	4.449	4.51	.817	-.082	.01369
9	.697	4.458	5.01	.896	-.079	.01817
10	.698	4.450	6.02	1.038	-.078	^a .03243
11	.695	4.442	7.03	1.117	-.075	.04912
12	.698	4.461	8.01	1.153	-.080	.07070

^aValue of c_d corrected for lost wake information.

^bThis point is a repeat of a previous one in this run. The angle of attack was approached from below.

^cThis is a hysteresis point. The angle of attack was approached from above.

TABLE III. Continued

(a) Continued

Point	M	R	α	c_n	c_m	c_d
Run 7						
1	0.800	7.834×10^6	-2.03	-0.029	-0.093	^a 0.01038
2	.797	7.796	-.98	.126	-.099	.00844
3	.798	7.794	.03	.277	-.105	^a .00862
4	.802	7.802	1.03	.426	-.110	^a .00984
5	.796	7.745	2.03	.575	-.111	.01105
6	.801	7.757	2.52	.640	-.119	^a .01522
7	.800	7.757	3.02	.694	-.118	^a .01926
8	.806	7.772	3.50	.728	-.124	^a .02770
Run 8						
2	0.762	7.837×10^6	-2.01	-0.008	-0.092	0.00811
3	.758	7.764	-.99	.132	-.094	.00771
4	.758	7.775	.01	.267	-.097	.00763
5	.756	7.694	1.00	.401	-.098	.00785
6	.757	7.706	2.00	.538	-.098	.00830
7	.758	7.759	3.01	.697	-.097	^a .01001
8	.756	7.725	3.50	.768	-.096	.01242
10	.755	7.728	4.50	.902	-.105	^a .02266
11	.762	7.836	5.00	.938	-.109	.03300
12	.765	7.732	6.02	.964	-.115	^a .06682
^c 14	.754	7.632	4.00	.841	-.103	
Run 9						
1	0.699	7.777×10^6	-2.02	-0.005	-0.085	
^b 2	.702	7.843	-2.06	-.007	-.086	0.00766
3	.699	7.775	.04	.251	-.090	.00757
4	.697	7.785	1.02	.373	-.091	.00767
5	.697	7.708	2.03	.499	-.092	.00781
6	.698	7.786	3.05	.627	-.090	.00839
7	.697	7.771	3.51	.683	-.089	.00904
8	.698	7.763	4.03	.754	-.086	.01076
9	.702	7.787	4.52	.837	-.083	.01399
10	.698	7.701	5.03	.900	-.080	.01830
11	.700	7.729	6.01	1.036	-.078	
^b 12	.701	7.776	6.02	1.042	-.080	
^b 13	.701	7.720	6.04	1.042	-.079	.03336
14	.699	7.782	7.02	1.110	-.075	^a .05600
^b 15	.700	7.779	7.04	1.109	-.076	.05146
16	.696	7.706	8.00	1.135	-.080	^a .07941

See footnotes on page 235.

TABLE III. Continued

(a) Continued

Point	M	R	α	c_n	c_m	c_d
Run 10						
1	0.760	7.923×10^6	0.00	0.268	-0.097	0.00791
2	.761	7.872	1.00	.406	-.099	.00818
3	.758	7.742	3.00	.684	-.095	a.00996
4	.757	7.684	3.51	.772	-.098	a.01276
5	.760	7.727	4.00	.847	-.104	a.01758
7	.760	7.822	4.51	.900	-.105	.02413
b ₈	.756	7.800	4.00	.846	-.102	.01733
b ₉	.758	7.807	3.49	.770	-.098	.01269
b ₁₀	.757	7.840	3.01	.700	-.097	a.00999
Run 11						
1	0.800	1.403×10^7	-2.14	-0.031	-0.098	0.00948
2	.798	1.398	-.99	.135	-.102	.00786
3	.797	1.390	.01	.283	-.106	.00785
4	.804	1.392	1.00	.437	-.114	.00936
5	.798	1.372	2.00	.587	-.116	.01060
6	.806	1.385	2.50	.643	-.124	.01704
7	.806	1.385	3.02	.690	-.124	.02206
8	.796	1.373	3.51	.773	-.127	.02278
9	.802	1.379	4.02	.782	-.127	
Run 12						
1	0.755	1.397×10^7	-2.02	-0.011	-0.094	0.00733
2	.757	1.404	.00	.278	-.099	.00717
3	.758	1.403	1.00	.414	-.100	.00747
4	.760	1.405	2.01	.551	-.101	.00784
5	.757	1.389	3.01	.695	-.098	.00948
6	.762	1.397	3.54	.787	-.102	a.01300
7	.761	1.407	4.02	.845	-.104	a.01705
8	.755	1.393	4.53	.905	-.105	a.02266
9	.766	1.410	5.01	.952	-.116	.03364
10	.760	1.391	6.03	.980	-.110	a.06281
11	.759	1.400	7.04	.973	-.108	

See footnotes on page 235.

TABLE III. Continued

(a) Concluded

Point	M	R	α	c_n	c_m	c_d
Run 13						
2	0.703	1.419×10^7	-2.01	-0.008	-0.089	0.00707
3	.692	1.378	.05	.251	-.093	.00703
4	.702	1.423	1.07	.383	-.094	.00721
5	.698	1.407	2.05	.506	-.094	.00740
6	.706	1.417	3.07	.638	-.093	.00790
7	.699	1.408	3.51	.694	-.090	.00855
8	.699	1.399	4.04	.766	-.088	.01017
9	.702	1.398	4.50	.844	-.086	.01349
10	.700	1.405	5.02	.907	-.082	.01816
11	.699	1.405	6.03	1.040	-.079	.03238
12	.702	1.412	7.05	1.126	-.079	.05261
13	.701	1.430	8.04	1.086	-.090	.08067
Run 14						
1	0.755	3.010×10^7	-2.02	-0.009	-0.098	0.00655
2	.759	3.017	-2.02	-.008	-.099	.00668
3	.760	3.022	.03	.290	-.104	.00646
4	.756	3.010	1.03	.428	-.104	.00658
5	.757	2.998	2.04	.566	-.104	.00704
6	.760	3.007	3.04	.730	-.104	.00902
7	.757	3.007	3.53	.798	-.104	.01175
8	.756	3.001	4.04	.870	-.109	.01711
9	.763	3.026	4.57	.938	-.122	.02511
10	.760	3.010	5.03	.973	-.119	.03307
Run 15						
11	0.754	3.000×10^7	6.04	1.016	-0.118	0.06643
Run 16						
1	0.764	2.992×10^7	3.93	0.890	-0.117	^a 0.01661
2	.759	2.980	6.03	1.006	-.126	.07106
Run 17						
1	0.749	2.893×10^7	4.04	0.865	-0.107	0.01648
2	.759	2.954	4.04	.894	-.118	^a 0.01680
3	.762	2.963	6.04	.970	-.121	.07232

See footnotes on page 235.

TABLE III. Continued

(b) Free transition

Point	M	R	α	c_n	c_m	c_d
Run 18						
1	0.802	4.480×10^6	-2.05	-0.020	-0.098	0.00835
2	.803	4.479	-.99	.148	-.109	.00631
3	.802	4.478	.03	.295	-.113	^a .00689
4	.800	4.468	1.02	.435	-.115	^a .00826
5	.802	4.459	2.05	.614	-.129	.01056
6	.799	4.442	2.09	.600	-.118	^a .00926
7	.799	4.441	2.53	.670	-.124	.01136
8	.803	4.454	3.02	.719	-.128	.01702
9	.800	4.435	3.52	.781	-.135	^a .02100
10	.803	4.440	4.02	.813	-.138	^a .02967
Run 19						
1	0.779	4.415×10^6	2.00	0.006	-0.099	0.00738
2	.783	4.454	-.99	.153	-.105	.00636
3	.781	4.444	.01	.294	-.107	.00600
4	.781	4.443	.99	.427	-.108	.00652
^a 5	.780	4.432	2.01	.572	-.106	^a .00718
^a 6	.779	4.415	3.03	.731	-.110	^a .00957
^a 7	.785	4.439	3.52	.796	-.119	^a .01466
^a 8	.777	4.410	4.01	.845	-.117	^a .01835
^a 9	.779	4.412	4.50	.889	-.120	^a .02511
^a 10	.778	4.397	5.01	.930	-.123	^a .03376
^a 11	.779	4.406	6.01	1.014	-.139	^a .05607
Run 20						
1	0.759	4.423×10^6	-2.01	0.009	-0.097	0.00722
2	.758	4.417	.00	.285	-.103	.00618
3	.758	4.433	1.00	.413	-.103	.00692
4	.760	4.426	2.00	.541	-.100	^a .00741
5	.759	4.433	2.01	.540	-.101	.00722
6	.761	4.466	3.01	.698	-.099	^a .00954
7	.762	4.431	3.50	.780	-.102	^a .01048
8	.764	4.441	4.00	.852	-.107	^a .01568
9	.757	4.411	4.51	.900	-.106	^a .02250
^b 10	.765	4.440	4.52	.900	-.110	^a .02288
11	.762	4.421	5.02	.937	-.108	^a .03095
12	.755	4.400	6.02	1.018	-.107	.04978
13	.760	4.417	7.02	1.085	-.117	.07357

See footnotes on page 235.

TABLE III. Continued

(b) Continued

Point	M	R	α	c_n	c_m	c_d
Run 21						
1	0.739	4.472×10^6	-2.03	0.007	-0.095	0.00721
2	.736	4.478	.00	.276	-.100	.00658
3	.737	4.483	1.00	.402	-.100	.00654
4	.739	4.470	2.00	.527	-.099	.00698
5	.741	4.445	3.00	.666	-.095	.00843
6	.738	4.431	3.49	.738	-.094	.01044
7	.739	4.444	4.03	.828	-.094	^a .01470
8	.737	4.424	4.49	.898	-.096	.02001
9	.737	4.419	5.00	.952	-.095	.02735
10	.746	4.459	6.03	1.001	-.099	^a .04579
^b 11	.738	4.460	6.03	1.038	-.098	.04398
12	.738	4.432	7.01	1.098	-.100	.06162
Run 22						
13	0.745	4.419×10^6	8.02	1.156	-0.109	0.08576
Run 23						
1	0.701	4.525×10^6	-2.01	0.021	-0.092	0.00714
2	.699	4.470	.00	.266	-.095	.00672
3	.697	4.461	1.01	.384	-.095	.00650
4	.700	4.505	2.00	.507	-.096	.00689
5	.698	4.492	3.00	.629	-.090	.00778
6	.700	4.474	3.50	.683	-.089	.00861
7	.699	4.464	4.00	.748	-.087	.00967
8	.700	4.474	4.50	.832	-.084	.01240
9	.699	4.458	5.00	.899	-.082	.01698
10	.698	4.452	6.03	1.049	-.081	.03115
11	.700	4.462	7.00	1.128	-.081	.04887
12	.701	4.485	7.99	1.170	-.082	.06418
13	.700	4.478	9.05	1.178	-.087	.07533

See footnotes on page 235.

TABLE III. Continued

(b) Continued

Point	M	R	α	c_n	c_m	c_d
Run 24						
1	0.603	4.516×10^6	-2.12	-0.012	-0.079	0.00786
3	.601	4.491	-.02	.229	-.084	.00609
4	.601	4.490	1.02	.345	-.085	.00635
5	.602	4.491	2.04	.456	-.085	.00679
6	.601	4.480	3.05	.563	-.084	.00746
7	.605	4.490	3.52	.617	-.084	.00780
8	.601	4.474	4.03	.673	-.083	.00809
9	.601	4.468	4.51	.726	-.081	.00858
10	.602	4.464	5.02	.766	-.078	.00960
11	.603	4.460	6.01	.880	-.070	.01445
12	.601	4.445	7.02	1.005	-.060	.02351
13	.601	4.443	8.01	1.061	-.051	.03601
14	.601	4.452	9.04	1.052	-.055	.05453
Run 25						
1	0.404	4.498×10^6	-2.00	0.016	-0.070	0.00750
2	.400	4.524	-.01	.213	-.073	.00691
3	.401	4.535	1.01	.317	-.075	.00681
4	.403	4.549	2.00	.412	-.077	.00693
5	.402	4.541	3.00	.502	-.075	.00730
6	.400	4.528	3.49	.551	-.075	.00763
7	.402	4.545	4.00	.600	-.076	.00775
8	.402	4.551	4.50	.643	-.075	.00810
9	.401	4.522	4.99	.692	-.075	.00827
10	.403	4.552	.01	.789	-.075	.00896
11	.400	4.527	7.00	.879	-.075	.00994
12	.400	4.526	8.00	.952	-.071	.01307
13	.402	4.541	8.99	.989	-.060	.02488
14	.401	4.540	10.01	1.006	-.051	.03833
Run 26						
1	0.799	7.843×10^6	-2.09	-0.035	-0.091	0.00933
2	.802	7.817	-1.00	.128	-.099	.00858
3	.798	7.823	.00	.274	-.103	.00842
4	.800	7.820	1.00	.416	-.106	.00893
5	.801	7.786	2.01	.563	-.108	.01125
^b 6	.798	7.662	2.52	.658	-.125	.01497
7	.802	7.761	2.53	.643	-.115	.01404
8	.801	7.718	3.01	.702	-.117	.01719
9	.804	7.748	3.49	.737	-.123	^a .02322
^b 10	.800	7.757	3.50	.757	-.123	.02158
11	.798	7.721	4.01	.805	-.124	.02824

See footnotes on page 235.

TABLE III. Continued

(b) Continued

Point	M	R	α	c_n	c_m	c_d
Run 27						
1	0.779	7.824×10^6	-2.05	-0.019	-0.092	
2	.779	7.822	-2.05	-.021	-.092	0.00824
3	.781	7.838	-1.01	.129	-.096	.00812
4	.779	7.819	-.00	.268	-.098	.00802
5	.782	7.833	1.00	.407	-.101	.00836
6	.781	7.783	2.00	.554	-.101	.00870
7	.779	7.742	3.00	.703	-.103	.01076
8	.780	7.747	3.00	.704	-.103	.01110
9	.783	7.750	3.49	.769	-.109	^a .01527
10	.781	7.741	4.00	.825	-.112	.02038
11	.783	7.736	4.49	.867	-.116	.02839
12	.780	7.724	5.00	.882	-.109	.03988
Run 28						
1	0.759	7.835×10^6	-2.02	-0.009	-0.091	0.00792
2	.756	7.846	-.01	.263	-.095	.00775
3	.760	7.841	.99	.397	-.097	.00806
4	.760	7.833	2.00	.533	-.097	.00835
5	.760	7.823	3.00	.678	-.094	.01022
6	.760	7.798	3.49	.760	-.096	.01261
7	.763	7.794	4.00	.824	-.099	.01696
8	.758	7.758	4.49	.877	-.098	.02431
9	.763	7.730	5.00	.914	-.103	.03392
10	.756	7.691	6.02	.991	-.103	.05504
^c 11	.754	7.635	5.00	.919	-.099	.03253
^c 12	.759	7.684	4.51	.879	-.100	.02394
^c 13	.759	7.740	4.00	.825	-.099	.01689
^c 14	.763	7.764	3.51	.757	-.095	^a .01227
^c 15	.763	7.726	2.00	.680	-.095	
^c 16	.752	7.314	.99	.395	-.098	.00795

See footnotes on page 235.

TABLE III. Continued

(b) Continued

Point	M	R	α	c_n	c_m	c_d
Run 29						
1	0.740	7.790×10^6	-2.05	-0.010	-0.089	0.00778
2	.737	7.783	.00	.261	-.093	.00760
3	.738	7.798	1.00	.389	-.095	.00781
4	.733	7.799	1.99	.512	-.093	.00805
5	.743	7.779	3.01	.654	-.092	.00964
6	.740	7.731	3.50	.730	-.090	.01159
8	.743	7.797	4.00	.826	-.093	.01677
9	.740	7.780	4.50	.891	-.091	.02186
10	.742	7.789	5.00	.939	-.092	.02992
11	.744	7.791	6.01	1.005	-.094	.04705
12	.741	7.733	7.01	.995	-.102	.08168
Run 30						
1	0.756	7.813×10^6	0.00	0.264	-0.096	0.00777
2	.758	7.837	2.01	.534	-.097	.00836
3	.756	7.642	3.50	.761	-.097	^a .01271
4	.758	7.725	3.99	.823	-.097	.01690
5	.756	7.751	4.50	.888	-.101	.02424
b_6	.762	7.781	2.01	.534	-.097	.00837
b_7	.762	7.792	3.49	.764	-.097	.01264
b_8	.761	7.790	4.00	.833	-.100	.01706
b_9	.762	7.795	4.49	.892	-.105	.02365
Run 31						
1	0.700	7.827×10^6	-2.01	-0.014	-0.085	0.00756
2	.700	7.836	.06	.243	-.089	.00759
3	.700	7.827	1.04	.365	-.090	.00780
4	.702	7.843	2.04	.488	-.090	.00795
5	.700	7.786	3.08	.619	-.089	.00842
6	.699	7.775	3.55	.681	-.086	.00896
7	.697	7.745	4.04	.730	-.084	.01025
8	.699	7.771	4.53	.814	-.082	.01294
9	.707	7.746	5.02	.909	-.081	.01892
10	.700	7.757	6.04	1.017	-.075	.03158
11	.700	7.721	7.03	1.102	-.076	.05126
12	.701	7.719	8.01	1.085	-.093	.08795

See footnotes on page 235.

TABLE III. Continued

(b) Continued

Point	M	R	α	c_n	c_m	c_d
Run 32						
1	0.604	7.754×10^6	-2.01	0.006	-0.080	0.00733
2	.603	7.771	-.00	.238	-.082	.00730
3	.602	7.787	.99	.348	-.083	.00744
4	.603	7.798	1.99	.458	.084	.00762
5	.601	7.752	3.00	.569	-.084	.00791
6	.603	7.791	3.50	.623	-.083	.00817
7	.601	7.779	4.00	.680	-.083	.00837
8	.599	7.732	4.51	.732	-.081	.00861
9	.600	7.729	5.00	.783	-.078	.00960
10	.603	7.787	6.01	.893	-.071	.01507
11	.600	7.726	7.01	.985	-.063	.02502
^b 12	.603	7.796	7.04	.988	-.064	.02532
13	.607	7.817	8.00	1.068	-.054	.04054
14	.602	7.780	9.01	1.030	-.056	.05489
Run 33						
1	0.804	1.399×10^7	-2.01	-0.017	-0.098	^a 0.00907
2	.801	1.396	-.98	.136	-.102	.00772
3	.797	1.390	.02	.282	-.106	.00776
4	.799	1.393	1.00	.428	-.110	.00826
5	.804	1.395	2.00	.581	-.117	^a .01230
6	.804	1.391	2.50	.650	-.121	.01444
7	.806	1.391	3.01	.708	-.125	.01990
8	.801	1.385	3.51	.770	-.128	.02341
9	.805	1.376	4.09	.750	-.122	^a .03298
Run 34						
1	0.779	1.416×10^7	-2.01	-0.032	-0.095	0.00750
2	.773	1.405	-1.00	.128	-.098	.00724
3	.778	1.416	.02	.270	-.102	.00726
4	.778	1.408	1.03	.411	-.103	.00750
5	.775	1.397	2.04	.551	-.102	.00800
6	.777	1.387	3.05	.705	-.103	.01010
7	.776	1.388	3.52	.768	-.105	.01408
8	.778	1.389	4.08	.837	-.115	^a .02055
9	.786	1.460	4.52	.906	-.130	.03211
10	.779	1.417	5.02	.906	-.118	.03808
11	.769	1.381	6.05	.919	-.110	.07040

See footnotes on page 235.

TABLE III. Continued

(b) Continued

Point	M	R	α	c_n	c_m	c_d
Run 35 (repeat of run 33)						
1	0.803	1.405×10^7	-0.99	0.134	-0.103	0.00787
2	.800	1.403	.01	.281	-.107	.00778
3	.804	1.384	2.02	.587	-.121	^a .01217
^b 4	.803	1.397	2.04	.591	-.119	.01179
5	.801	1.392	2.50	.654	-.120	.01386
^b 6	.802	1.393	2.52	.653	-.120	.01457
7	.797	1.399	1.00	.425	-.109	.00818
Run 36						
1	0.760	1.416×10^7	-2.01	-0.011	-0.094	0.00735
2	.757	1.403	.01	.263	-.098	.00719
3	.761	1.407	1.01	.404	-.101	.00744
4	.761	1.404	2.00	.538	-.100	.00774
5	.768	1.407	3.04	.699	-.098	^a .00959
6	.759	1.398	3.51	.764	-.097	.01216
7	.762	1.401	4.03	.843	-.105	^a .01673
8	.764	1.402	4.53	.884	-.105	.02303
9	.761	1.404	5.02	.934	-.104	.03157
10	.762	1.403	6.03	.957	-.113	.05901
Run 37						
1	0.742	1.408×10^7	-2.02	-0.011	-0.092	0.00718
2	.739	1.405	-.01	.262	-.096	.00707
3	.743	1.407	1.00	.395	-.097	.00724
4	.740	1.402	2.02	.524	-.097	.00749
5	.741	1.401	3.01	.663	-.093	.00892
6	.740	1.398	3.51	.737	-.092	.01120
7	.742	1.388	4.03	.826	-.094	.01602
8	.738	1.411	4.49	.872	-.090	.02121
9	.744	1.403	5.01	.933	-.095	.02999
10	.740	1.394	6.03	1.025	-.095	.04569
11	.743	1.399	7.02	.999	-.102	.07602

See footnotes on page 235.

TABLE III. Continued

(b) Continued

Point	M	R	α	c_n	c_m	c_d
Run 38						
1	0.702	1.413×10^7	-2.00	-0.003	-0.089	0.00699
2	.703	1.417	-.00	.245	-.092	.00701
3	.703	1.417	1.02	.380	-.094	.00709
4	.703	1.415	2.03	.501	-.093	.00730
5	.701	1.419	3.01	.632	-.091	.00786
6	.697	1.402	3.50	.690	-.088	.00841
7	.710	1.406	4.03	.770	-.087	.01086
8	.709	1.428	4.50	.842	-.086	.01426
9	.704	1.406	5.00	.904	-.080	
b_{10}	.706	1.405	5.01	.917	-.082	.01916
11	.701	1.399	6.02	1.041	-.079	.03260
12	.704	1.410	7.04	1.112	-.078	.05286
Run 39						
1	0.606	1.429×10^7	-2.03	0.002	-0.082	
b_2	.601	1.406	-2.03	.004	-.082	0.00682
3	.603	1.401	-.01	.240	-.085	.00670
4	.601	1.402	1.00	.353	-.085	.00683
5	.604	1.419	2.01	.469	-.087	.00697
6	.602	1.411	2.99	.580	-.086	.00735
7	.603	1.409	3.49	.641	-.087	.00754
8	.602	1.410	4.00	.699	-.087	.00775
10	.602	1.421	4.49	.751	-.084	.00810
11	.606	1.414	5.00	.803	-.081	.00949
12	.604	1.409	6.00	.921	-.073	.01506
13	.601	1.389	7.00	.995	-.064	.02490
14	.606	1.425	8.00	1.039	-.054	.04507
Run 40						
1	0.800	3.009×10^7	-2.08	-0.020	-0.102	0.00813
2	.800	2.982	-.98	.143	-.107	.00681
3	.804	3.045	.05	.306	-.116	^a .00728
4	.801	3.027	1.02	.447	-.117	.00792
5	.801	2.992	2.04	.593	-.122	^a .01130
6	.801	2.992	2.54	.669	-.127	.01481
8	.801	2.997	3.01	.718	-.131	.01972
9	.805	3.010	3.53	.756	-.131	.02780

See footnotes on page 235.

TABLE III. Continued

(b) Continued

Point	M	R	α	c_n	c_m	c_d
Run 41						
1	0.780	2.983×10^7	-2.03	-0.014	-0.100	0.00680
2	.780	2.987	-.98	.147	-.104	.00651
3	.785	3.003	.03	.294	-.108	.00669
4	.781	2.979	1.03	.435	-.109	.00671
5	.781	2.999	2.04	.595	-.111	.00722
6	.783	3.010	3.03	.743	-.118	.01255
7	.783	2.988	3.53	.791	-.120	^a .01626
8	.781	3.004	4.03	.859	-.125	.02220
9	.783	3.003	4.53	.892	-.130	.03191
10	.780	2.990	5.04	.898	-.126	.04301
Run 43						
1	0.763	3.039×10^7	-2.01	-0.002	-0.098	0.00653
2	.760	3.024	.01	.283	-.103	.00640
3	.758	3.013	1.01	.420	-.104	.00651
4	.761	3.003	2.04	.562	-.104	.00700
5	.762	2.987	3.03	.727	-.103	^a .00876
6	.763	2.997	3.53	.816	-.111	^a .01213
7	.759	2.974	4.02	.864	-.110	^a .01560
8	.762	2.996	4.53	.916	-.113	.02384
9	.765	2.976	5.01	.961	-.120	.03312
10	.760	2.994	6.04	.998	-.120	^a .06696
^c 11	.760	2.986	4.96	.968	-.117	.03070
^c 12	.760	2.980	4.47	.915	-.112	^a .02269
^b 13	.759	3.030	.01	.281	-.103	
^b 14	.760	3.010	.01	.283	-.103	.00637
^b 15	.762	3.030	1.02	.425	-.106	.00661
^b 16	.762	3.029	2.04	.564	-.105	.00694
^b 17	.759	3.000	3.02	.728	-.104	.00881
Run 44						
1	0.743	2.999×10^7	-2.13	-0.014	-0.096	0.00645
2	.739	2.999	.02	.279	-.100	.00632
3	.738	3.000	1.03	.414	-.101	.00633
4	.743	3.000	2.04	.550	-.102	.00677
5	.741	3.004	3.03	.696	-.098	.00854
6	.741	3.000	3.52	.780	-.098	.01196
7	.743	2.983	4.04	.857	-.098	.01605
8	.740	2.984	4.53	.924	-.100	.02234
9	.742	2.960	5.03	.991	-.105	.03238
10	.741	2.973	6.03	1.046	-.104	.04957

See footnotes on page 235.

TABLE III. Continued

(b) Continued

Point	M	R	α	c_n	c_m	c_d
Run 45						
1	0.702	3.021×10^7	-2.00	0.009	-0.092	0.00629
2	.702	3.008	.03	.270	-.096	.00625
3	.702	3.019	1.03	.399	-.096	.00635
4	.701	3.004	2.04	.526	-.097	.00657
5	.704	3.039	3.03	.660	-.094	.00706
6	.702	2.997	3.53	.706	-.093	.00801
7	.701	3.005	4.01	.787	-.089	.00963
8	.704	3.035	4.51	.869	.087	^a .01388
9	.704	2.995	5.01	.944	-.085	.01947
10	.704	2.989	6.02	1.087	-.085	.03501
11	.703	3.006	7.03	1.150	-.085	.05298
12	.707	3.009	8.05	1.092	-.098	^a .08398
Run 46						
1	0.404	3.025×10^7	-2.00	0.011	-0.075	0.00609
2	.400	3.000	.06	.229	-.078	.00596
3	.399	2.989	1.06	.332	-.078	.00619
4	.401	2.986	2.04	.435	-.079	.00641
5	.401	2.994	3.00	.531	-.080	.00671
6	.401	3.008	3.52	.586	-.080	.00676
7	.399	2.981	4.01	.636	-.081	.00714
8	.402	3.009	4.51	.688	-.081	.00728
9	.401	2.996	5.02	.740	-.081	.00746
10	.399	2.958	6.00	.836	-.080	.00809
11	.400	2.969	7.02	.939	-.080	.00910
12	.399	2.961	8.02	1.004	-.073	.01401
13	.401	2.983	9.00	1.038	-.057	.04130
Run 47						
1	0.599	2.981×10^7	-2.02	0.012	-0.084	0.00625
2	.600	3.000	-.01	.255	-.088	.00602
3	.601	3.011	1.00	.372	-.088	.00618
4	.600	3.013	2.01	.487	-.089	.00639
5	.601	3.019	3.00	.602	-.089	.00684
6	.600	3.016	3.51	.660	-.089	.00707
7	.602	3.027	3.98	.715	-.088	.00726
8	.601	3.000	4.50	.771	-.086	.00805
9	.599	3.014	5.00	.818	-.083	.00891
10	.601	3.003	6.02	.950	-.073	^a .01496
11	.603	3.015	7.01	1.024	-.066	^a .02512
12	.598	3.008	8.02	1.045	-.055	^a .04282

See footnotes on page 235.

TABLE III. Continued

(b) Continued

Point	M	R	α	c_n	c_m	c_d
Run 48						
1	0.802	5.010×10^7	-1.99	0.002	-0.106	0.00760
2	.800	4.991	-.97	.158	-.112	.00626
3	.801	5.021	.00	.314	-.117	.00651
4	.802	5.009	1.00	.468	-.123	^a .00785
Run 49						
^b 2	0.806	4.537×10^7	-2.12	-0.023	-0.104	^a 0.00897
3	.806	4.532	-1.99	.000	-.106	.00825
4	.804	4.519	-.99	.157	-.112	.00668
5	.803	4.518	.02	.312	-.117	^a .00659
6	.803	4.531	1.02	.468	-.123	^a .00837
Run 50						
1	0.701	4.491×10^7	-2.06	0.004	-0.093	0.00589
2	.701	4.499	.02	.274	-.097	.00583
3	.702	4.511	1.02	.405	-.098	.00595
4	.702	4.508	2.04	.536	-.098	.00608
5	.704	4.520	3.00	.669	-.096	.00657
6	.701	4.498	3.49	.714	-.094	.00734
7	.700	4.485	4.00	.799	-.091	.00932
8	.701	4.479	4.52	.874	-.088	^a .01279
9	.699	4.488	5.01	.953	-.085	^a .01849
10	.707	4.523	6.01	1.112	-.092	
Run 51						
1	0.803	4.467×10^7	2.03	0.611	-0.132	^a 0.01321
2	.799	4.478	2.54	.702	-.137	.01590
Run 52						
1	0.760	4.501×10^7	-2.02	-0.000	-0.100	0.00613
2	.762	4.489	.02	.294	-.106	.00590
3	.762	4.513	1.02	.437	-.107	.00615
4	.761	4.495	2.02	.577	-.107	.00639
5	.761	4.492	3.02	.735	-.103	.00810
6	.756	4.491	3.53	.825	-.110	.01127
7	.770	4.524	4.02	.905	-.129	^a .02043

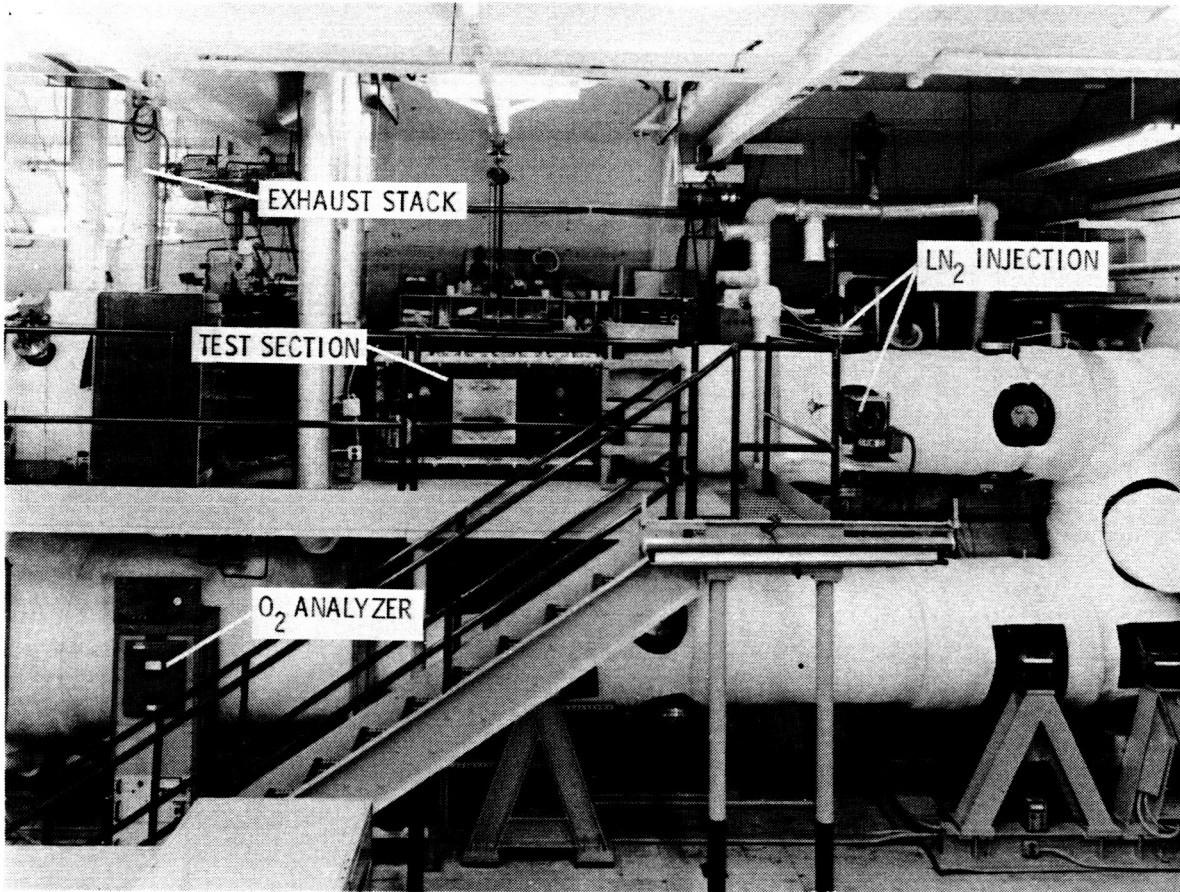
See footnotes on page 235.

TABLE III. Concluded

(b) Concluded

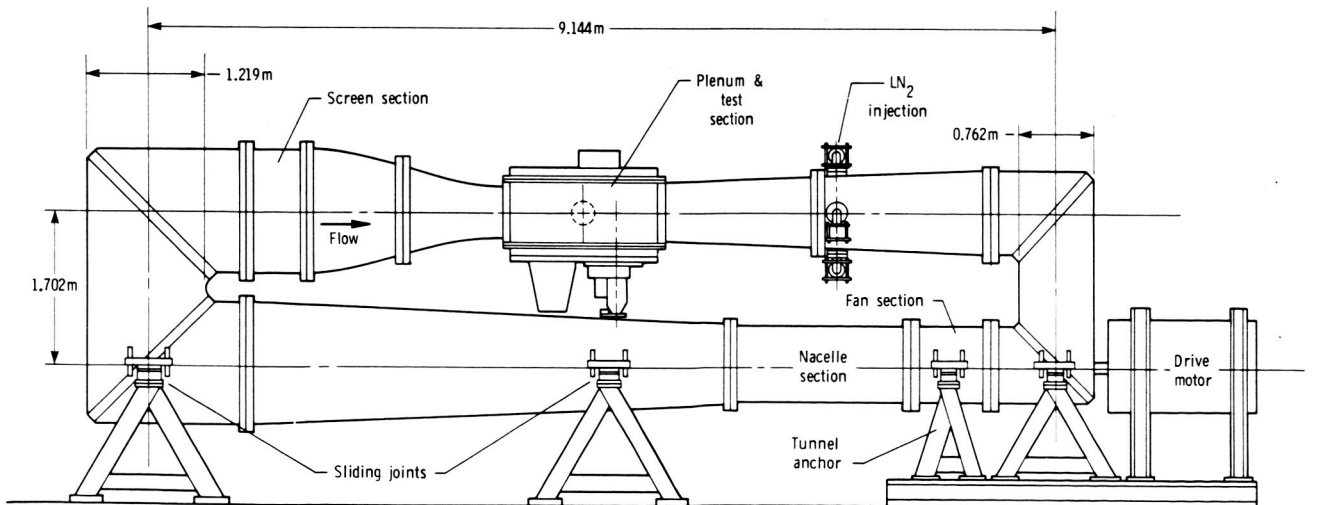
Point	M	R	α	c_n	c_m	c_d
Run 53						
1	0.784	4.483×10^7	-1.98	-0.002	-0.102	0.00642
2	.781	4.457	-.98	.156	-.107	.00626
3	.780	4.455	-.02	.300	-.110	.00614
4	.784	4.464	1.03	.454	-.113	.00652
5	.779	4.438	2.04	.603	-.111	.00683
6	.785	4.479	3.01	.779	-.131	^a .01468
^b 7	.784	4.463	3.01	.770	-.127	^a .01536
^b 8	.783	4.462	2.03	.612	-.116	.00754
9	.780	4.453	2.49	.686	-.114	.00929
Run 55						
1	0.737	4.484×10^7	0.01	0.287	-0.102	0.00589
2	.739	4.501	1.00	.423	-.103	.00606
3	.741	4.493	2.00	.559	-.103	.00618
4	.742	4.494	3.00	.720	-.101	.00803
5	.740	4.479	4.00	.868	-.101	.01579

See footnotes on page 235.



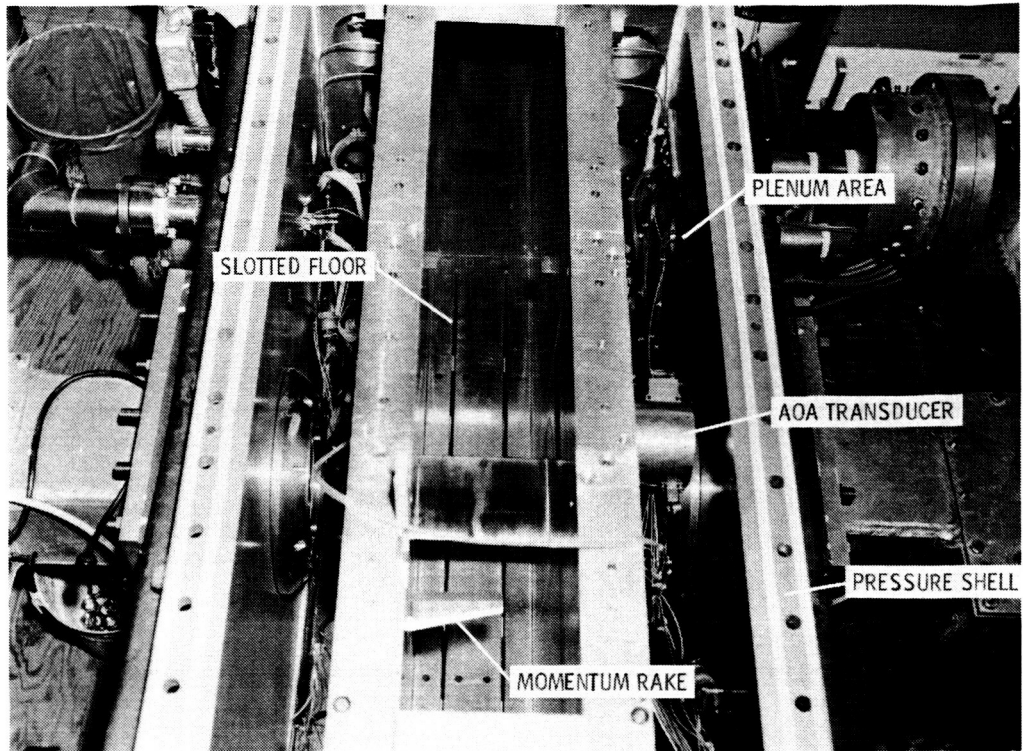
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(a) Photograph.



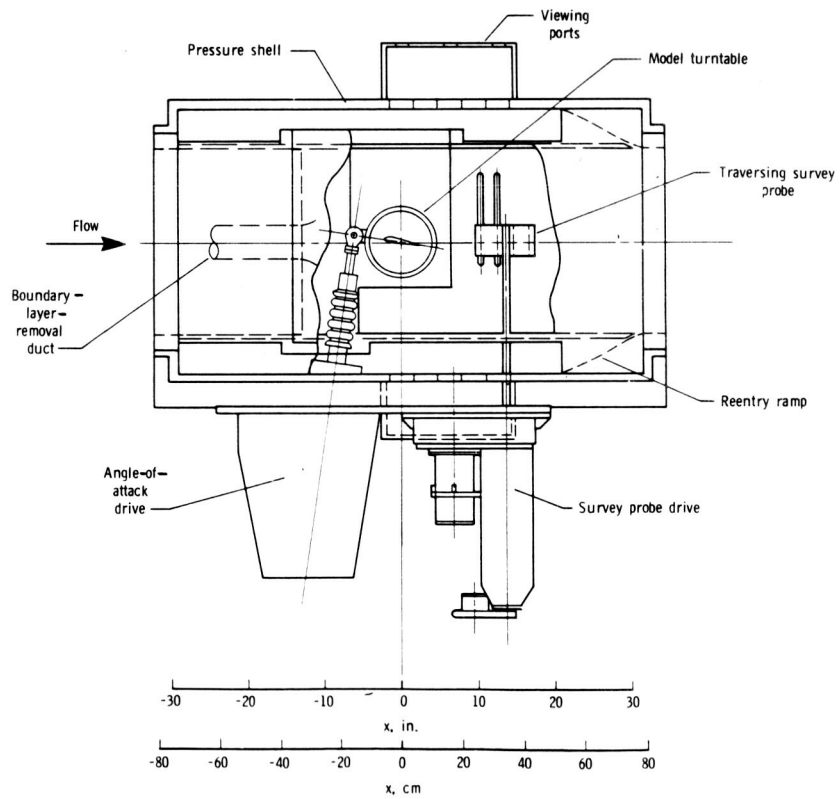
(b) Schematic drawing.

Figure 1. Elevation view of Langley 0.3-Meter Transonic Cryogenic Tunnel with two-dimensional test section installed.



L-79-8913.1

(a) Top-view photograph.



(b) Schematic drawing showing major components.

Figure 2. Two-dimensional test section.

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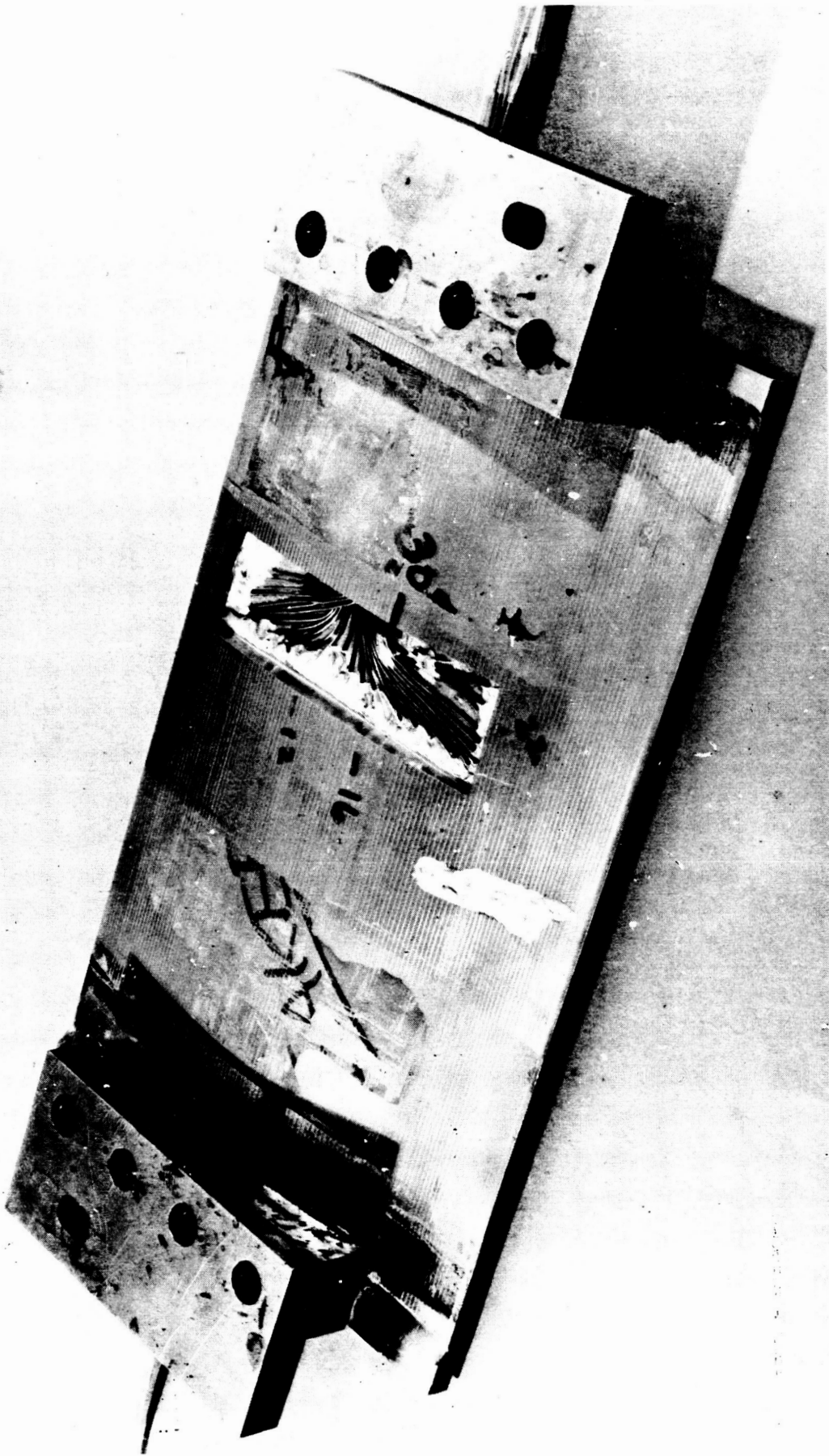


Figure 3. Model under construction.

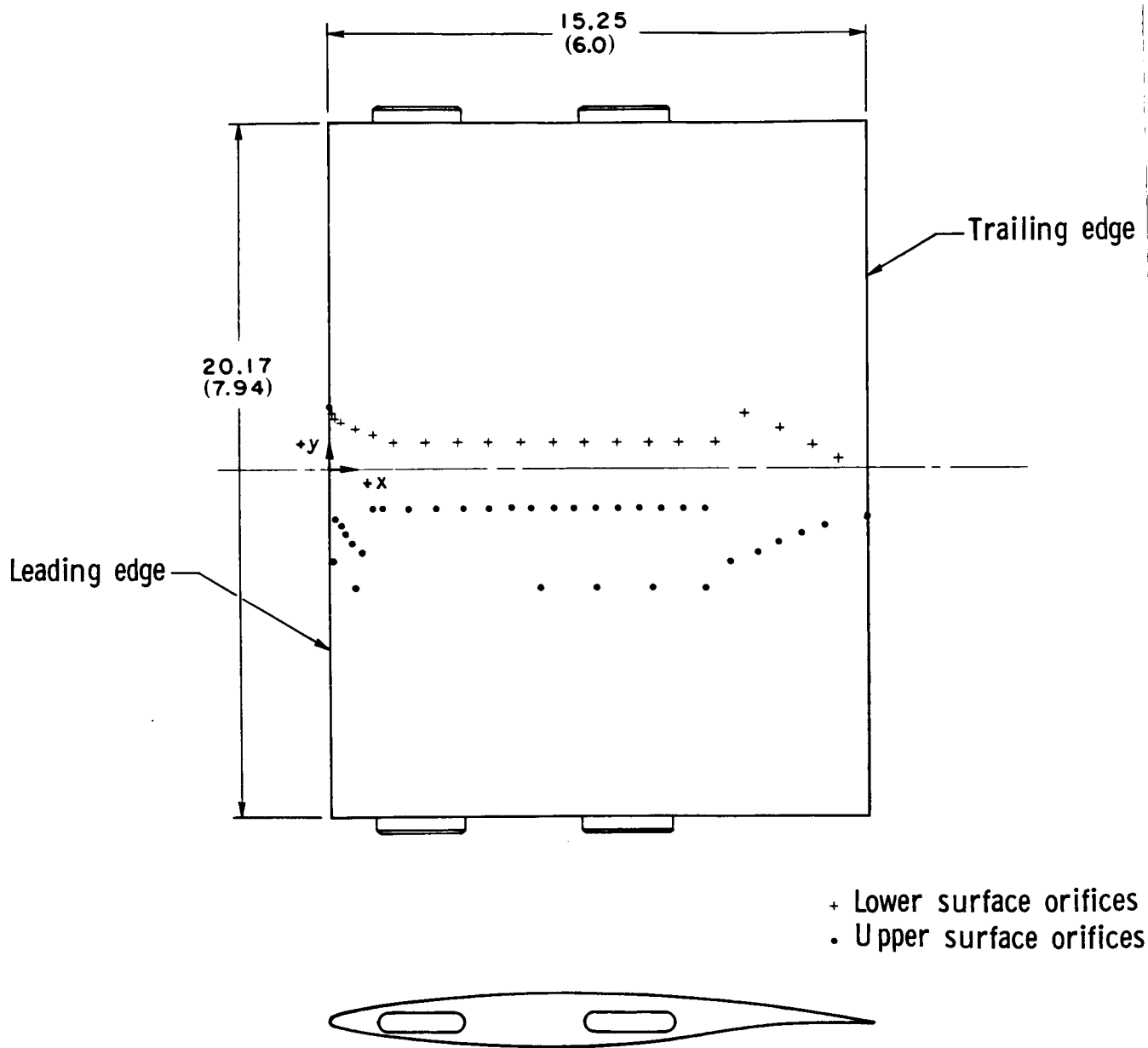
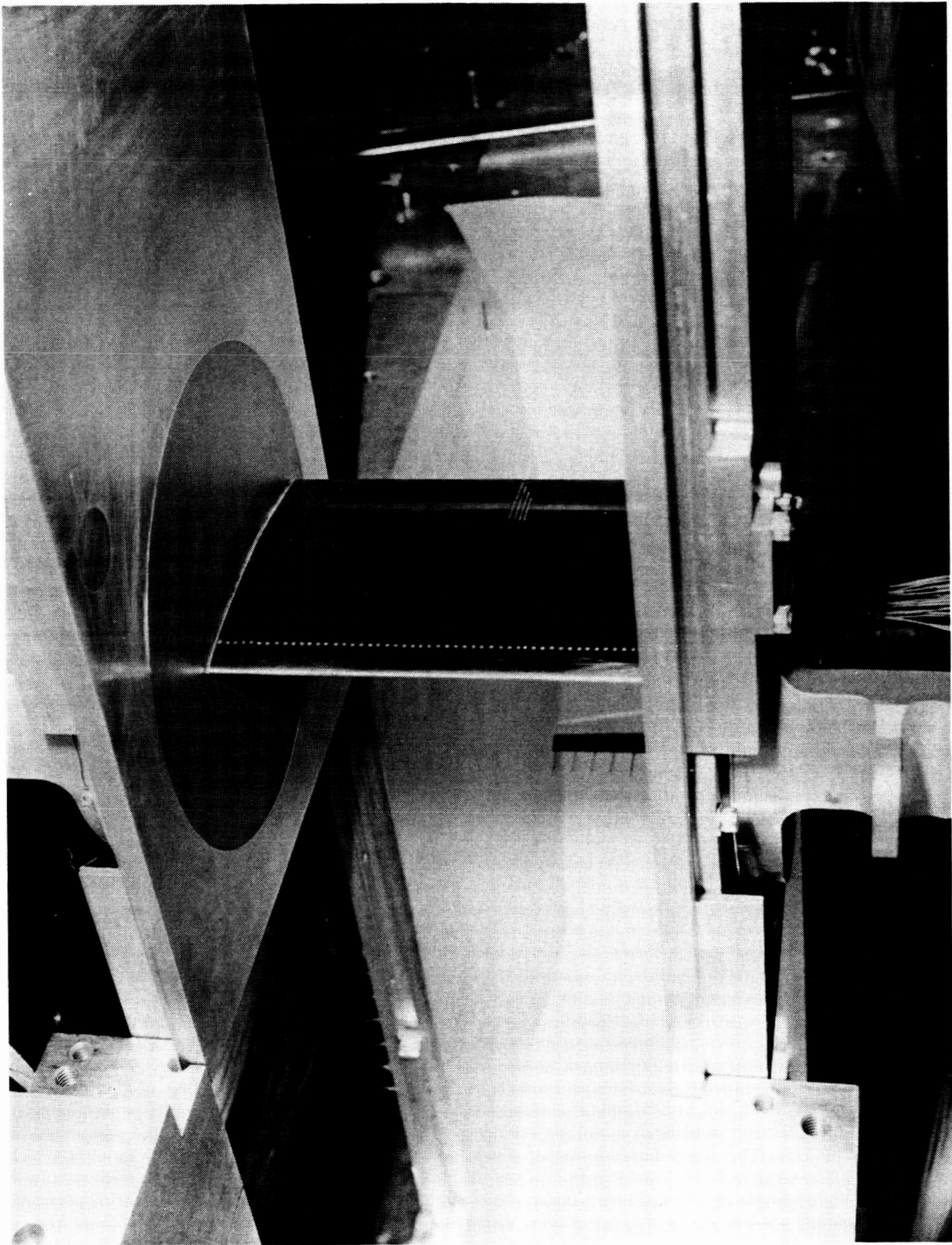


Figure 4. Schematic drawing of model showing orifice arrangement.



L-80-4438

Figure 5. Installation of airfoil model in tunnel test section.

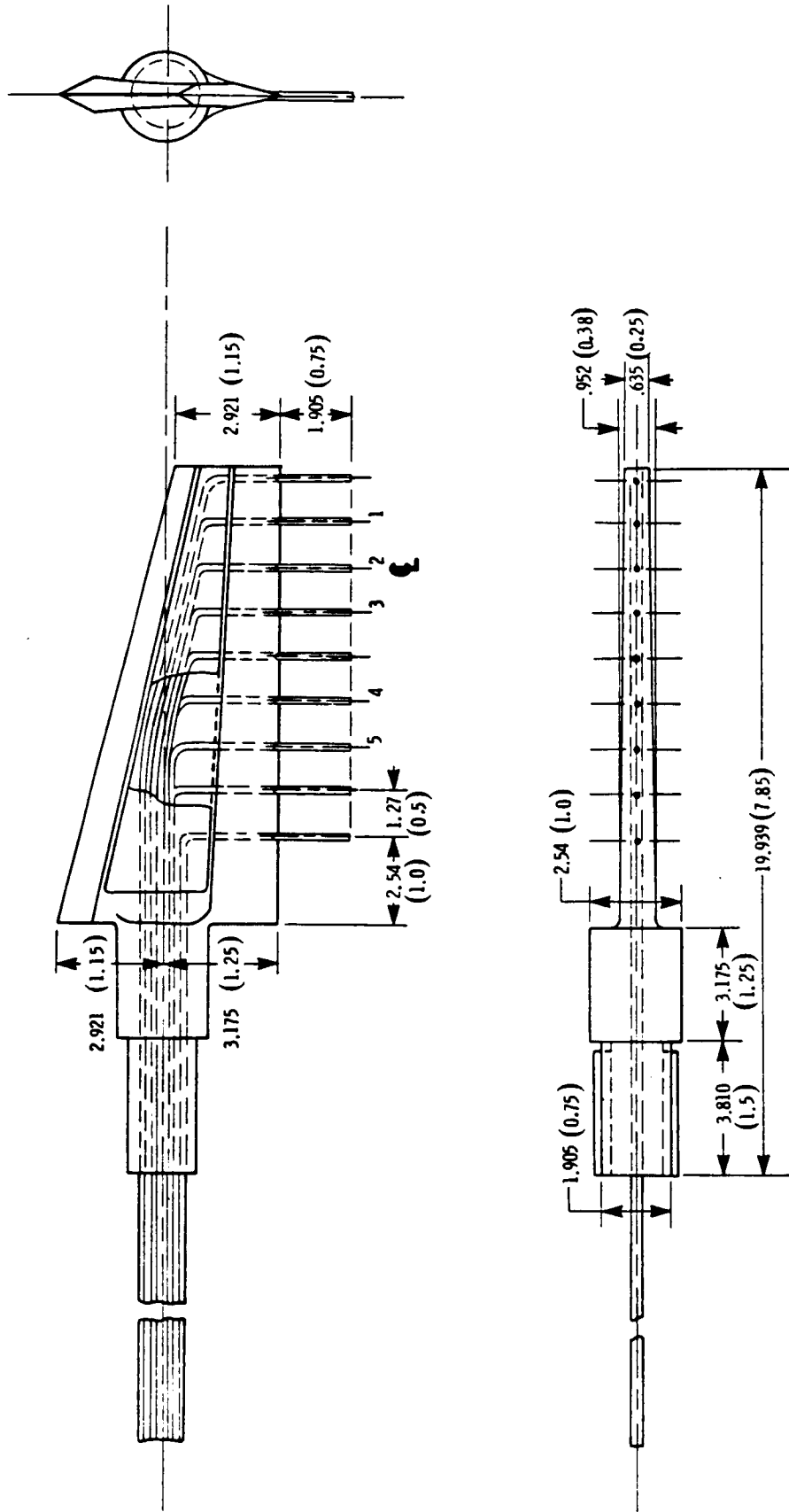


Figure 6. Details of wake survey probe. All dimensions are in centimeters (inches).

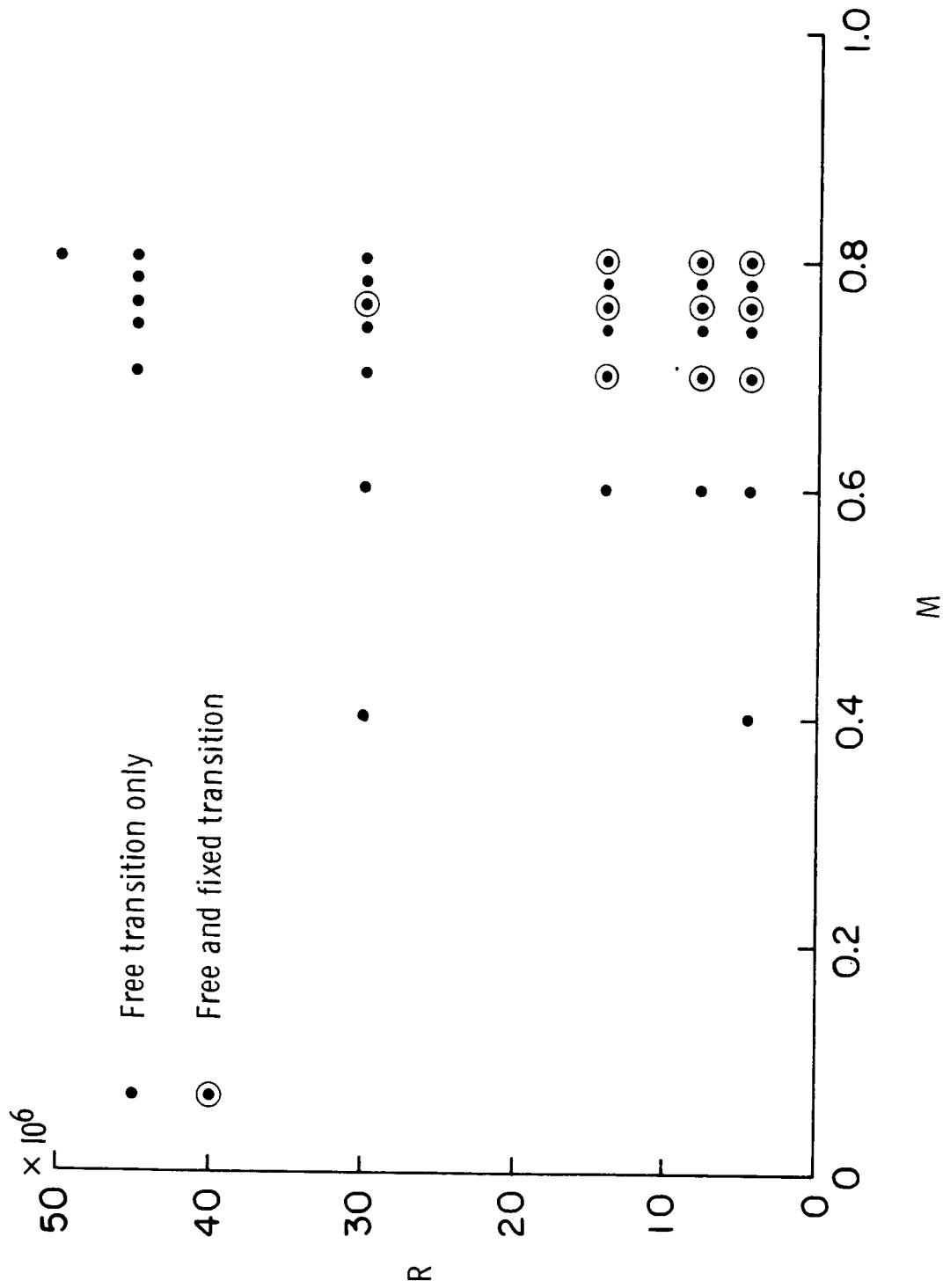


Figure 7. Range of Reynolds number and Mach number used in test program.

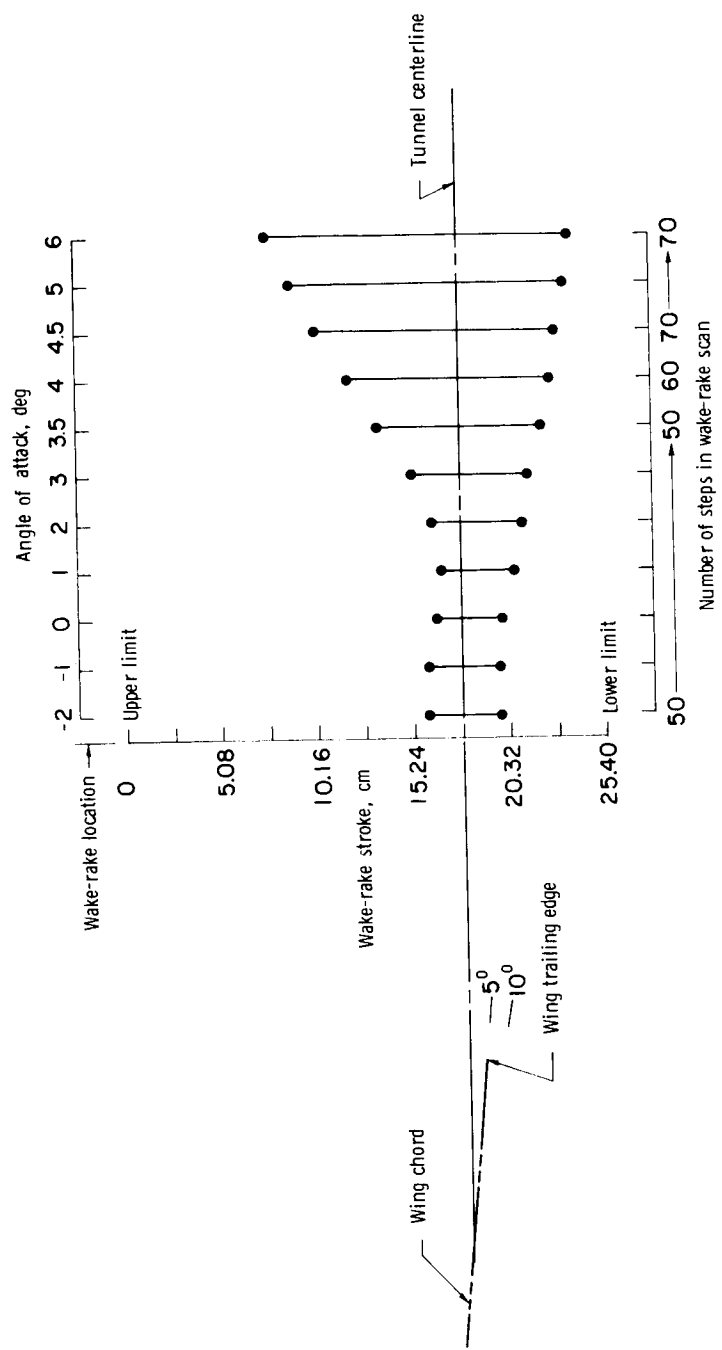
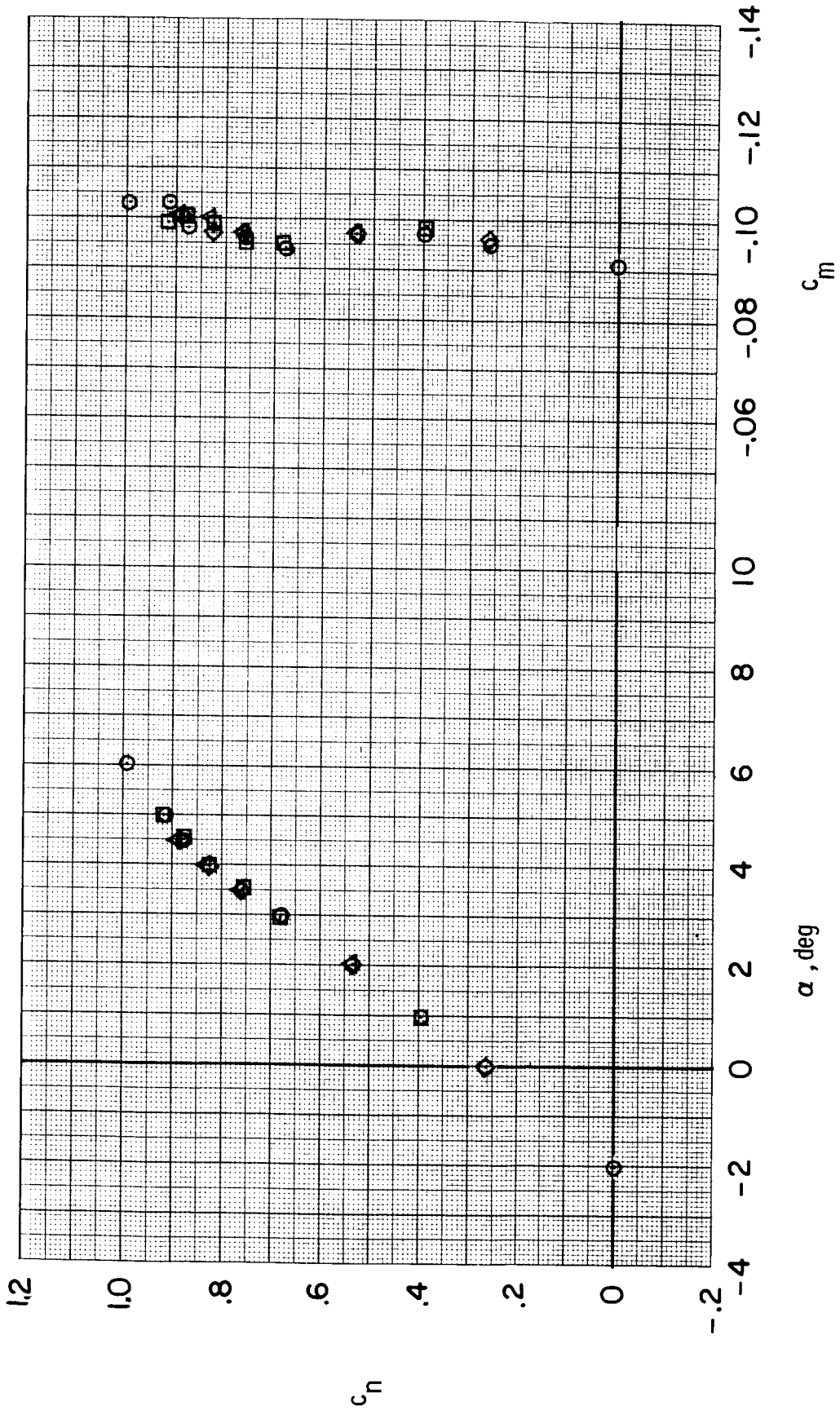
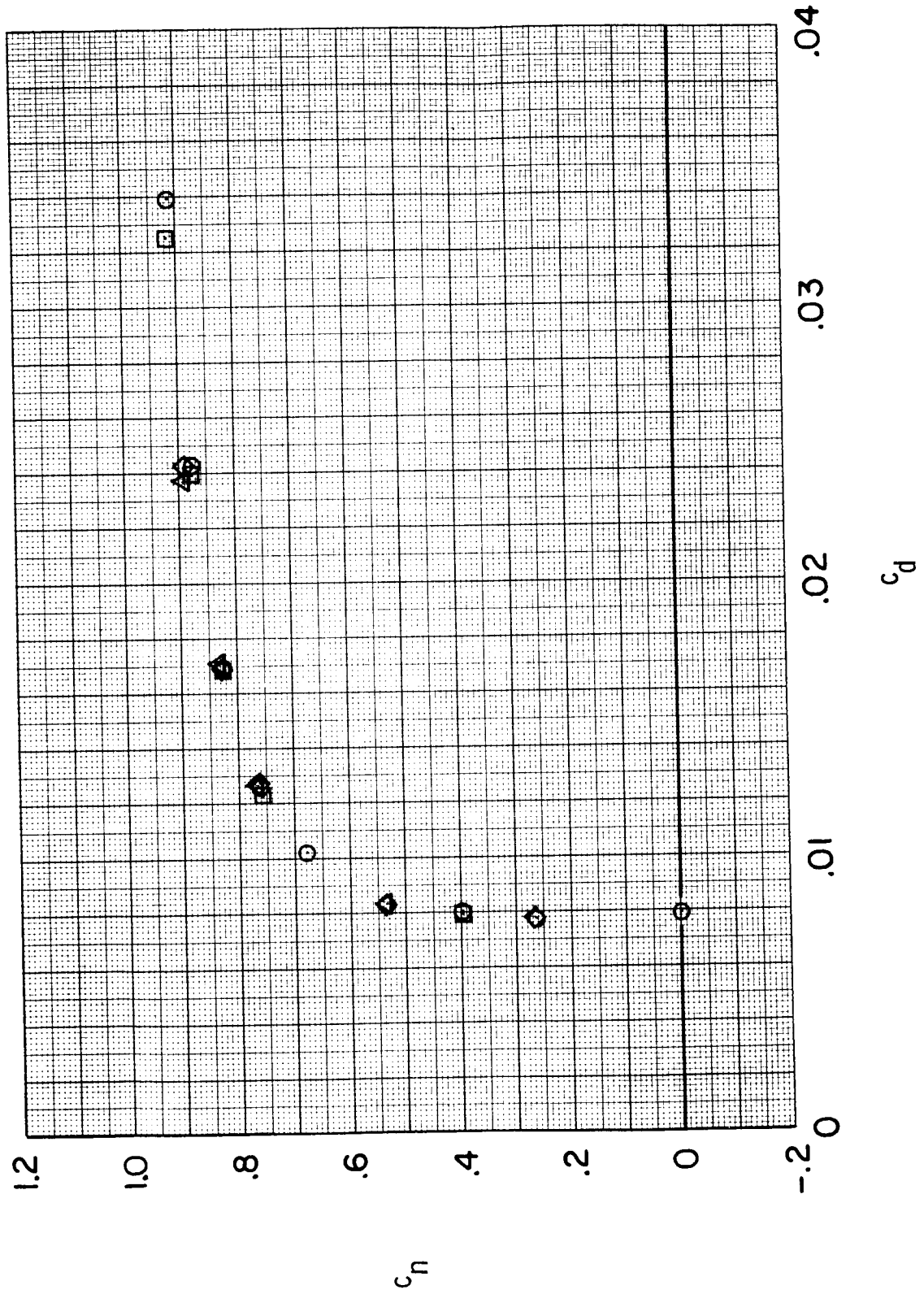


Figure 8. Variation of stroke length and number of steps used to define wake at $M \approx 0.76$.



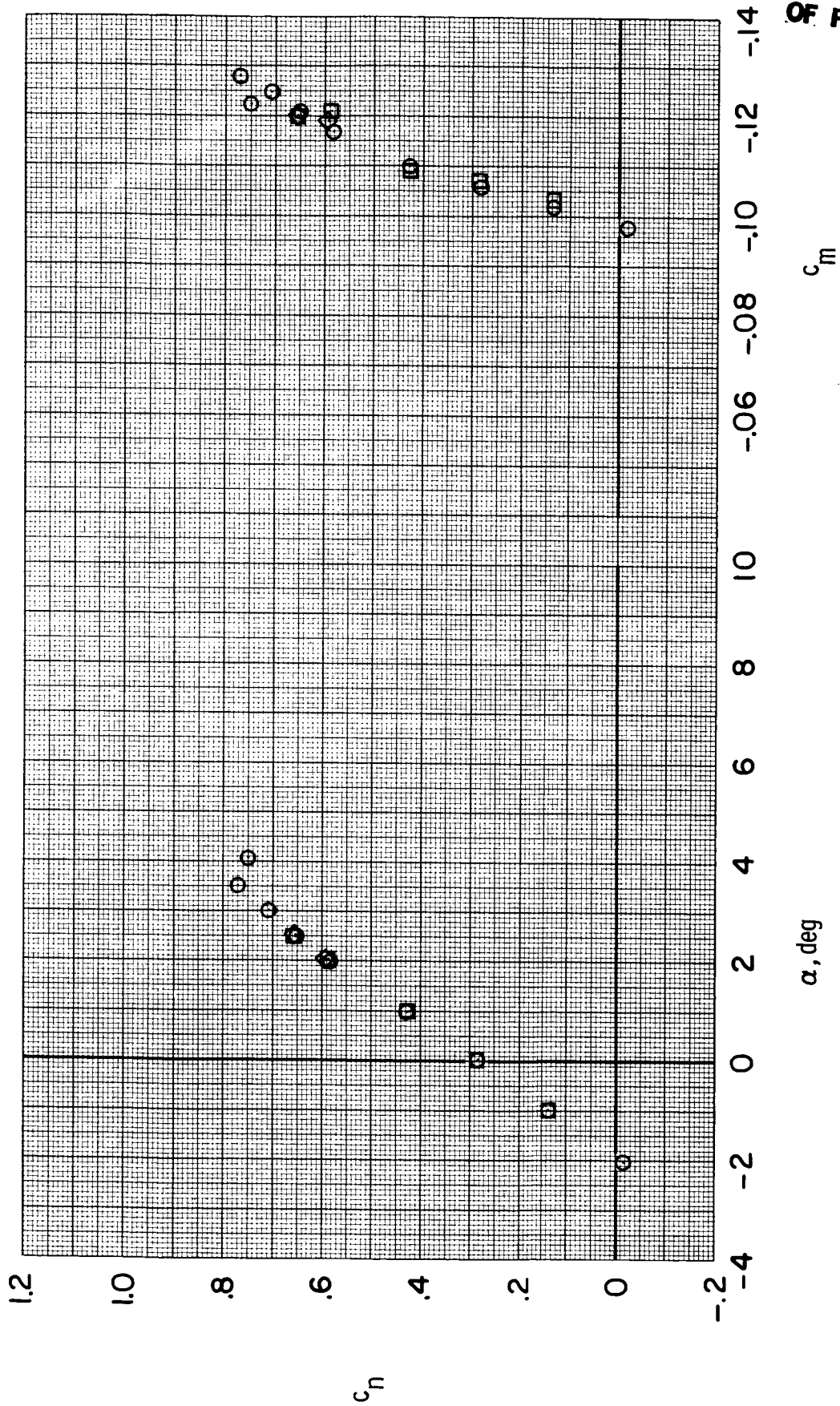
(a) c_n as function of α and c_m .

Figure 9. Repeatability of four sets of data ($\circ, \square, \diamond, \triangle$) with free transition at $M \approx 0.76$ and $R \approx 7.7 \times 10^6$.



(b) c_n as function of c_d .

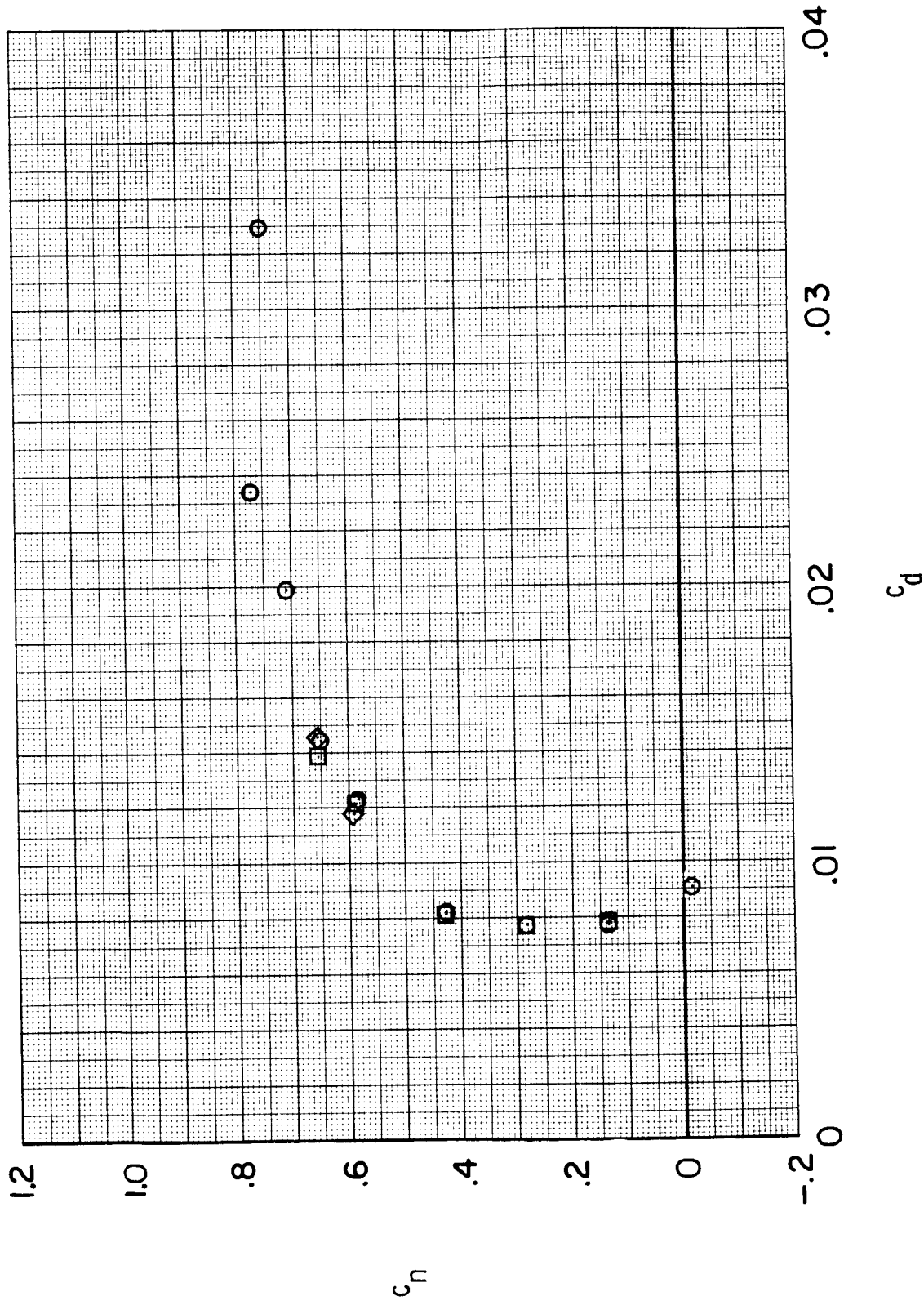
Figure 9. Concluded.



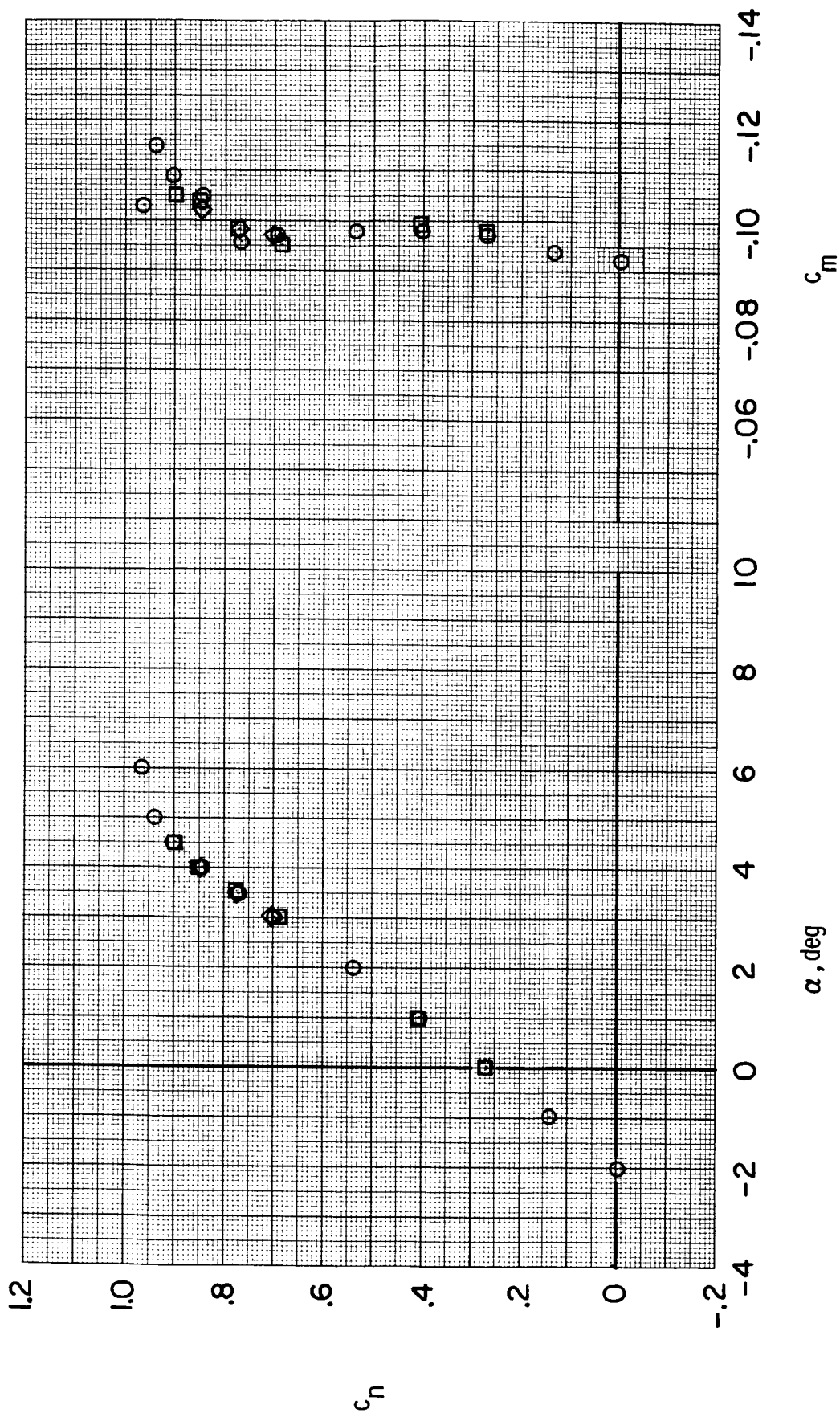
(a) c_n as function of α and c_m .

Figure 10. Repeatability of three sets of data (o, □, ◇) with free transition at $M \approx 0.80$ and $R \approx 14.0 \times 10^6$.

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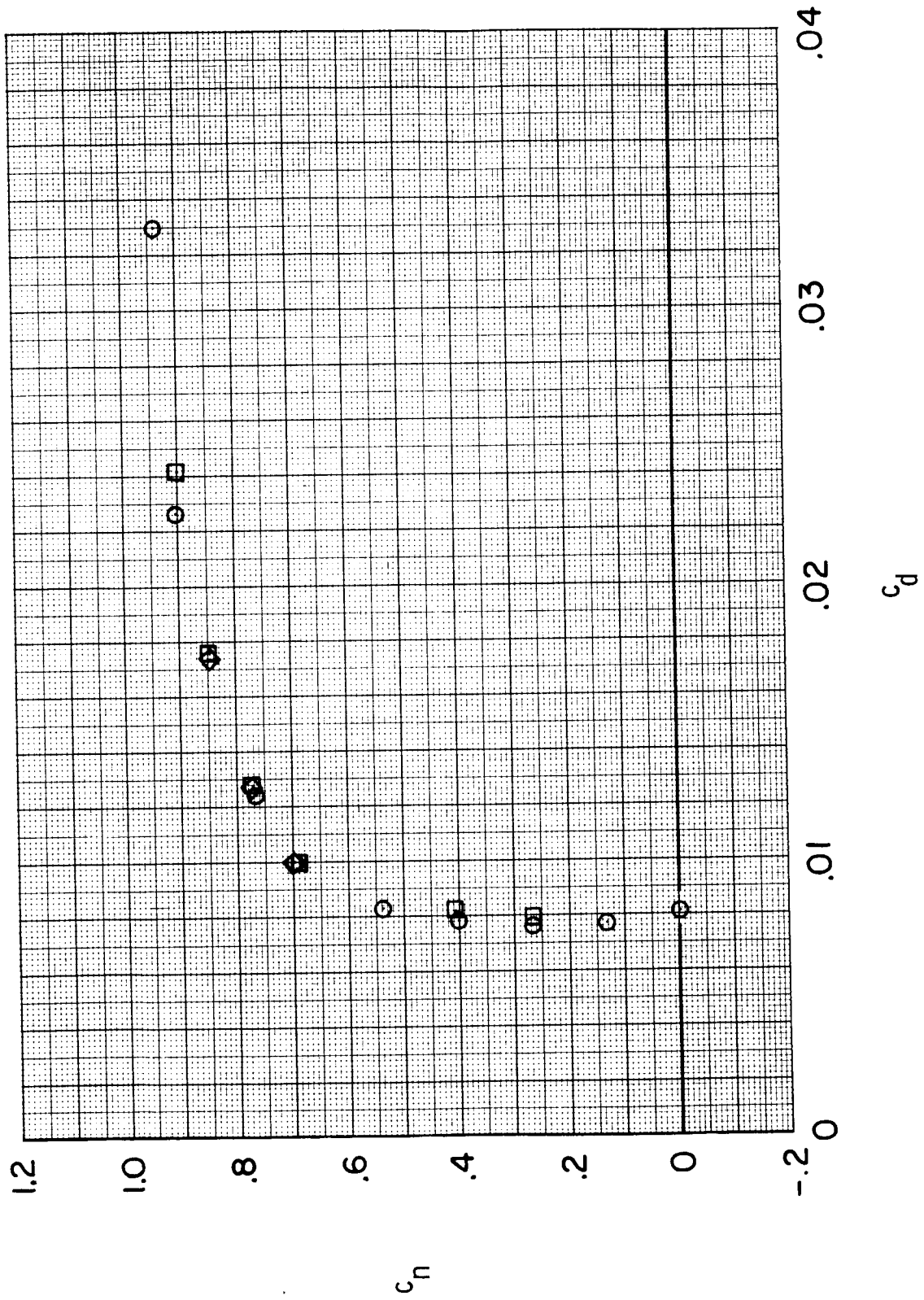


(b) c_n as function of c_d .
Figure 10. Concluded.



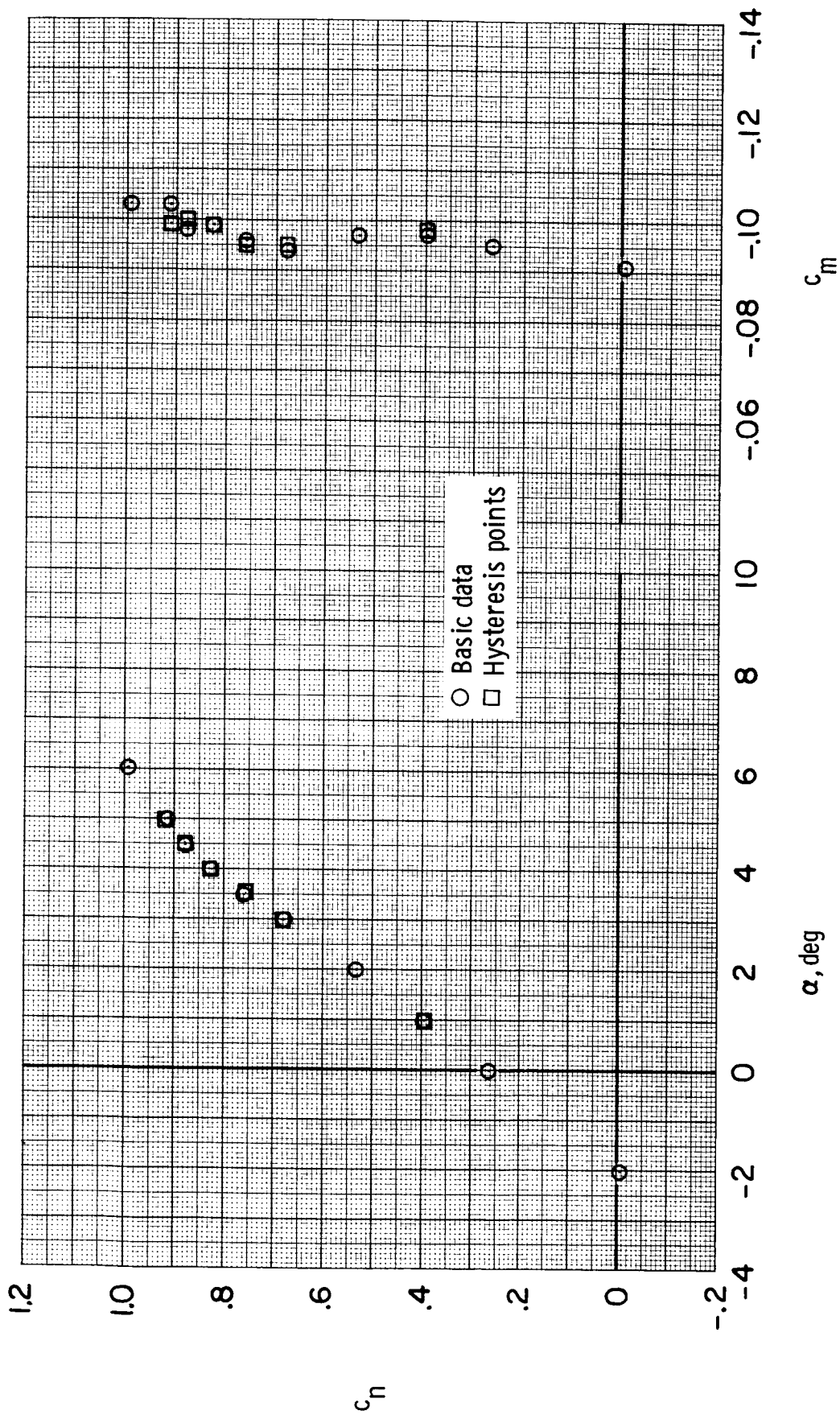
(a) c_n as function of α and c_m .

Figure 11. Repeatability of three sets of data (\circ , \square , \diamond) with fixed transition at $M \approx 0.76$ and $R \approx 7.7 \times 10^6$.



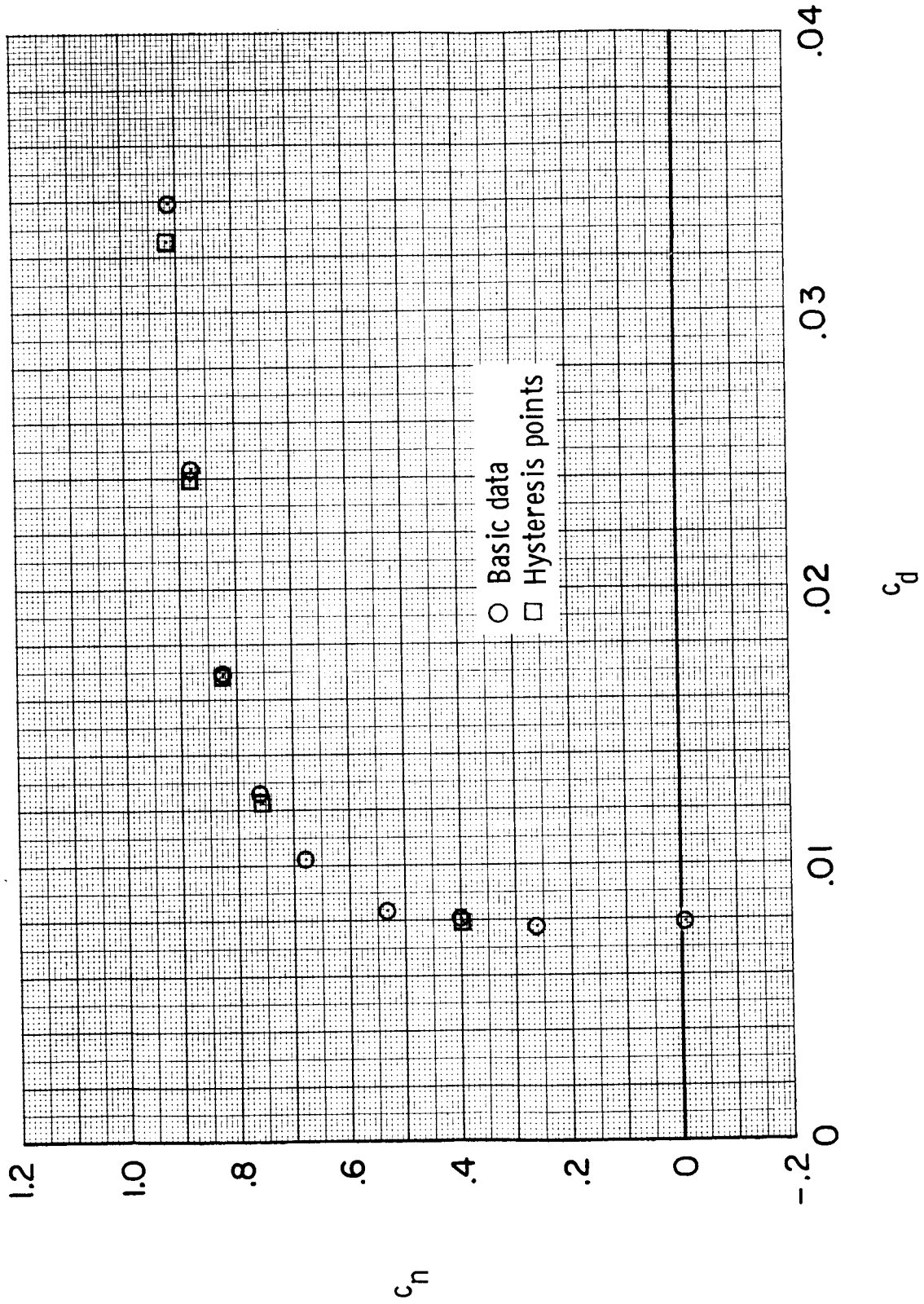
(b) c_n as function of c_d .

Figure 11. Concluded.



(a) c_n as function of α and c_m .

Figure 12. Hysteresis characteristics of data with free transition at $M \approx 0.76$ and $R \approx 7.7 \times 10^6$.



(b) c_n as function of c_d .
 Figure 12. Concluded.

Standard Bibliographic Page

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				15. Supplementary Notes	
16. Abstract A wind-tunnel investigation designed to test a Boeing advanced-technology airfoil from low to flight-equivalent Reynolds numbers has been completed in the Langley 0.3-Meter Transonic Cryogenic Tunnel. This investigation represents the first in a series of NASA/U.S. industry two-dimensional airfoil studies to be completed in the Advanced Technology Airfoil Test program. Test temperature was varied from ambient to about 100 K at pressures ranging from about 1.2 to 6.0 atm. Mach number was varied from about 0.40 to 0.80. These variables provided a Reynolds number (based on airfoil chord) range from 4.4×10^6 to 50.0×10^6 . All the test objectives were met. The pressure data are presented without analysis in plotted and tabulated formats for use in conjunction with the aerodynamic coefficient data published as NASA TM-81922. At the time of the test, these pressure data were considered proprietary and have only recently been made available by Boeing for general release. Data are included which demonstrate the effects of fixed transition. Also included are remarks on the model design, the model structural integrity, and the overall test experience.					
17. Key Words (Suggested by Authors(s)) Supercritical airfoil Two-dimensional airfoil Cryogenic wind tunnel High Reynolds number			18. Distribution Statement Unclassified—Unlimited Subject Category 02		
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