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NAVAL POSTGRADUATE SCHOOL

Monterey, California



THESIS

MARINE STEAM CONDENSER DESIGN OPTIMIZATION

by

Thomas M. Buckingham

December 1983

Thesis Advisor:

R. H. Nunn

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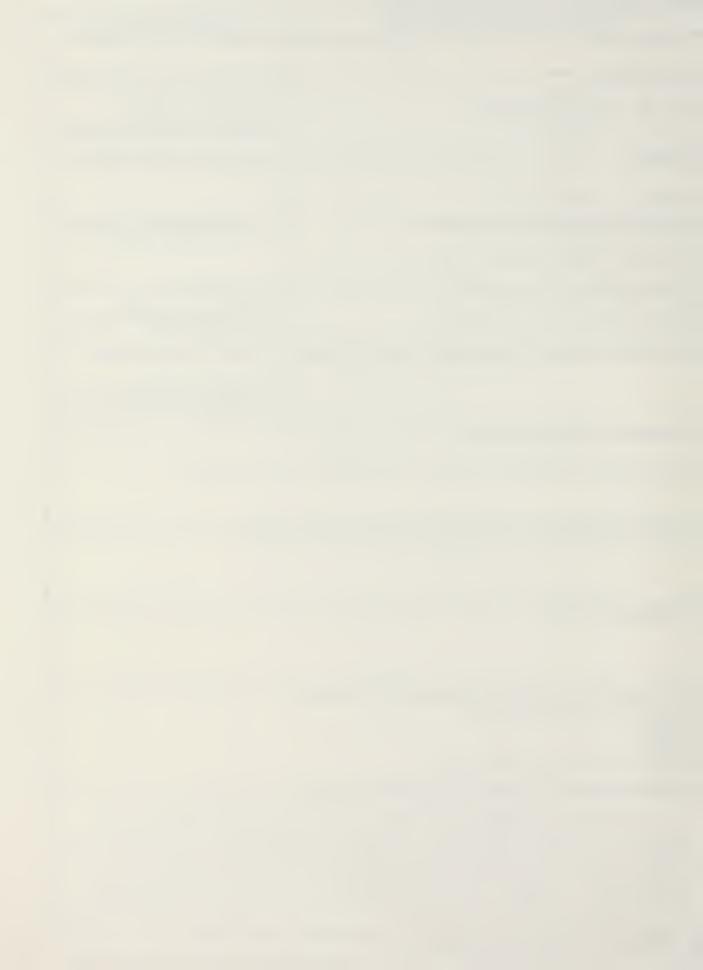
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CONDIP is an extremely versatile design tool, incorporating a detailed analysis of the complex steam-side thermodynamic processes occurring at each row in the condenser. The additional capability of tube enhancement is also included. However, in coupling CONDIP with CONMIN numerous problems had to be overcome in order to make CONDIP capable of completing an analysis even when thermodynamic conditions in the condenser became infeasible. This had to be accomplished while ensuring continuity in all constraint and objective function evaluations. A series of test cases were conducted to evaluate and compare the importance of various objective functions and design criteria.



Marine Steam Condenser Design Optimization

bу

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Submitted in partial fulfillment of the requirements for the degree of

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ABSTRACT

A surface-condenser analysis code was coupled with a constrained function minimization code to produce an automated marine condenser design and optimization package. The program, CONDIP, was based on the principles developed in ORCON1, a sophisticated computer code produced by the Oak Ridge National Laboratory. CONMIN, the optimization program, was developed at the Ames Research Center.

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I. INTRODUCTION

A. BACKROUND

For many years the steam plant was unchallenged in its role as the primary type of marine propulsion systems. But recently gas turbines have become a desirable alternative despite the fact that they are less efficient than comparable steam plants. The primary advantage of gas turbines is their light weight and compact size. Thus, in order for the marine steam plant to survive, it is imperative that lighter, more compact and efficient steam plants be developed.

while there are numerous advanced concepts in all areas of steam propulsion which can be explored, one simple way to streamline the steam plant is the elimination of overdesign. Most overdesign is due to unnecessary safety factors used to offset lack of detailed knowledge about the thermal processes in the plant. Identification of the minimum safe design could significantly reduce unnecessary overdesign and result in the development of a smaller, more compact power plant.

B. METHODOLOGY

In the United States the most prevalent criterion for the design and specification of surface condensers is based on the "square-root of V" relationship, as developed by the Heat Exhange Institute (HEI) [Ref. 1]. The HEI method was adopted by the Department of the Navy Bureau of Ships (now Naval Sea System Command) for specification of U.S. Naval condensers.



The HEI method is very simple in its approach, calculating the overall heat transfer coefficient as a function of cooling water velocity through the tubes, inlet coolant temprature, tube wall thickness and material, and fouling. The limitations of this method are apparent. Designs based on HEI are insensitive to shell-side conditions. Saturation steam pressure, temperature and enthalpy are assumed to be constant as steam passes through the bundle, whereas in reality there is a continual pressure drop as steam flow passes over the rows of tubes, with a corresponding decrease in saturation temperature. There is also no provision for any effects of condensate film, external tube enhancement, etc. on the shell-side of the bundle. In addition, the HEI method does not account for the presence and effect of non-condensible gases that inevitably contaminate a condenser.

With the capabilities of high speed computers now available, more comprehensive methods have been developed to account for the deficiencies of the HEI method. In particular, a radial flow computer code was developed to calculate the local heat transfer and thermodynamic properties on a row by row basis. Known as ORCON1, this code was developed by Oak Ridge National Laboratory (ORNL) under contract to the Office of Saline Water during the period from 1968-1970 [Ref. 2]. The program was based, in part, on the work performed by Eissenberg [Ref. 3]. Eissenberg's experimental results led to correction factors on the basic Nusselt equation to account for condensate inundation effects on tubes within a condenser bundle. Basically, ORCON1 divides the condenser into sectors and performs a row by row analysis within each sector, determing local heat transfer coefficients, heat flux, steam characteristics, the effect of condensate inundation and numerous other parameters at each row. ORCON1 is also capable of incorporating the effects of both tube-side and steam-side enhancement factors. Since



ORCON1 represented a much more comprehensive and detailed analysis of the condenser then the less exact HEI method, its results could be expected to be more precise.

Some work has been done at the Naval Postgraduate School to improve the capabilities of ORCON1. In his development of OPCODE2, Johnson [Ref. 5] added subroutines to ORCON1 which calculated tube-side pressure drops, corresponding pumping power and condenser volume. Nunn and Marto [Ref. 14] have incorporated the effects of vapor shear in an amended version of ORCON1 called MORCON. MORCON includes the correlations developed by Fujii [Ref. 4] to determine the effect of vapor velocity on the thermal resistance of the condensate film on the condenser tubes. In general, vapor shear effects tend to enhance the condenser heat transfer on the steam-side of the tube while condensate inundation tends to inhibit it.

The ability to represent numerically the actual thermodynamic processes occurring within the condenser has improved dramatically. However, the capability to couple these increasingly comprehensive and complex condenser design programs with an optimizing procedure has not made comparable progress. Optimization is a powerful tool which can help in reducing overdesign and achieving the goal of a safe compact condenser design.

There are currently numerous computer optimization programs available which can be coupled with general design programs of all types to numerically improve and ultimately determine the best design. The key is to properly write the design program so that it is compatible with the optimizer. Johnson [Ref. 5] developed a computer program called OPCODE1, based on the HEI method of condenser design, and was able to couple it with one such numerical optimizer. The results of OPCODE1 demonstrated how condenser designs can indeed be safely improved upon. It also revealed the



versatility of condenser design optimization as a powerful design tool. However, Johnson was unsuccessful in coupling OPCODE2 (his derivative of ORCON1) with an optimizer. This failure does not alter the fact that that in order to fully appreciate more sophisticated condenser design analyses, such as that used in ORCON1, it is imperative that computer programs be developed which will be compatible with current numerical optimizers.

C. OBJECTIVE

There were two primary objectives of this thesis. The first objective was to develop a computer code which incorporates the basic condenser analysis of ORCON1 and the subsequent improvements made in MORCON and OPCODE2, but which will be capable of being coupled with a numerical optimizer to yield a complete, detailed design package. This design package can then be used as a tool in obtaining a much more reasonable conceptual design and for use in comparison studies. It would provide the naval architect the ability to optimize weight, volume, cost or any other potential design objective of the marine plant.

The second objective was to make this design package capable of determining the single best design rather than simply an improvement over the initial design. The key was to construct the program in such a way so the optimizer does not stop at some relative optimum, but continues the analysis until no further improvement can be realized. It is most desirable to be able to reach this single true optimum design regardless of initial design variable values.



II. NUMERICAL OPTIMIZATION

A. BACKGROUND

Nearly all design problems require either the minimization or maximization of a parameter. This parameter will be called the problem's objective function or design objective [Ref. 6]. For a given design to be feasible or acceptable, it must satisfy a set of design constraints which are either maximum or minimum limiting values for a pre-determined set of parameters or functions of parameters. For example, in any condenser design the outer diameter of a condenser tube can never be less than zero and there is normally some practical upper limit which also cannot be exceeded. limits are design constraints on the tube outer diameter. In the design problem there is also a set of design variables which are parameters whose values can be changed specified limits in order to minimize or maximize the design objective. For example, in minimizing the condenser volume an engineer may want to vary tube inner diameter, tube wall thickness and tube length. These three parameters would thus be examples of typical design variables.

For such complex design problems as the treatment of the condenser design in ORCON1, it is necessary to choose an optimization scheme which can handle the problem and provide a rational, rapid approach to design automation and optimization. An optimization program based on direct methods for solution of constrained problems [Ref. 11] was chosen for this research work.



B. CONSTRAINED FUNCTION MINIMIZATION (CONMIN)

Vanderplaats [Ref. 7] developed an optimization program, CONMIN, capable of optimizing a very wide class of engineering problems. CONMIN is a fortran program, in subprogram form, that optimizes a multi-variable function subject to a set of inequality constraints.

It is practical at this point to introduce three basic definitions and their respective conditions [Ref. 8].

DESIGN VARIABLES: Those parameters which the optimization program is permitted to change in order to improve the design. Design variables appear only on the right side of an equation, are continuous, and have continuous first derivatives.

DESIGN CONSTRAINTS: Any parameter which must not exceed specified bounds for the design to be acceptable. Design constraints may be linear or nonlinear, implicit or explicit, but they must be functions of the design variables. Design constraints appear only on the left side of equations.

OBJECTIVE FUNCTION: The parameter which is going to be minimized or maximized during the optimization process. The objective function may also be either linear or nonlinear, implicit or explicit, and must be a function of the design variables. The objective function usually appears on the left side of an equation. The only exception is if the objective function is also a design variable.

Assuming that the optimization process requires the minimization of a particular objective function the general optimization problem can be stated as:

Find the vector of design variables, \underline{X} , To minimize the objective function, $F(\underline{X})$, Subject to the constraints:

$$VLB_1 \le X \le VUB_1$$
 $i = 1, NDV$ (eqn 2.2)

In the general problem, $G_j(\underline{X})$ are the constraint functions; there are NCON constraints and NDV design



variables; VLB_i and VUB_i are the lower bounds and upper bounds of the i-th design variable. If the equality condition is met, $(G_j(X)=0.)$, the constraint is active. If the inequality is met, $(G_j(X)<0.)$, the constraint is inactive. Finally, if the inequality of equation 2.1 is violated, $(G_j(X)>0.)$, that constraint is said to be violated. Because of numerical inaccuracies representing exact zero on the computer, the equality condition is represented by a band around the value $G_j(X)=(0.\pm CT)$ where CT is the constraint thickness.

Any design which satisfies the inequalities of equations 2.1 and 2.2, thus having no violated constraints, is said to be feasible. If the design violates any of these constraints it is said to be an infeasible design. The design which best mimimizes the objective function while still remaining feasible is said to be optimal.

, CONMIN requires an initial set of values for the design variables \underline{X} to obtain an initial design which is either feasible of infeasible. If the initial design is feasible, CONMIN moves in a direction which will minimize the objective function. If the initial design is infeasible, CONMIN moves toward a feasible solution with minimal increase in the object function.

The optimization process proceeds in an iterative fashion. Johnson [Ref. 5] presents in greater detail the procedures utilized in CONMIN to search for the minimum objective value. In general, the methods used by CONMIN to determine search direction include the method of steepest descent, the method of conjugate direction, and the method of feasible directions. For further background concerning CONMIN and the numerical techniques utilized in optimization, consult Vanderplaats [Ref. 7], Fletcher and Reeves [Ref. 9], Zoutendijk [Ref. 10], and Vanderplaats and Moses [Ref. 12]. However, it is necessary to stress a few pertinent points which will aid in understanding how the program was developed in this thesis.



The optimization process begins by calculating the gradient of the objective function using a finite difference technique. A perturbation is applied to each of the design variables in a single forward step and the gradient vector is determined.

$$\nabla_{E}(\overline{X}) = \begin{bmatrix} \frac{2X}{3E}(\overline{X}) \\ \frac{2X}{3E}(\overline{X}) \end{bmatrix}$$

The search direction is then calculated and is a function of this gradient and any active or violated constraints resulting from the applied perturbation. Subsequent search directions are a function of previous search directions, as well as current gradient information and any appropriate constraint factors. Obviously, the size of the perturbation and the size of the bandwidth about an active constraint will have a great deal of effect on the search direction and ultimate optimization process. This detail will be recalled later-on during the code development.

There are some limitations to CONMIN. The number of design variables (NDV) directly affects the computational time required to reach an optimum. Since the calculation of the gradient information required for each design variable at the beginning of each design iteration is found by using a single forward finite difference step, requiring a complete pass through the analysis portion of the program, there is an increase in CPU time as NDV increases. Also, as NDV increases, there is the corresponding rise in machine related numerical innacuracy. Vanderplaats [Ref. 6] recommends no more than twenty as a practical limit for the number of design variables.

It is quite possible that while design improvement may be obtained, the single best design optimum or true optimum



may not be reached. This is not an uncommon occurrence and there are several possible explanations. For example, the design problem may not be formulated properly or the analysis may be extremely complex and non-linear. However, a more common reason is that there are "relative optimums" between the initial design and the single true optimum. This concept of relative vs true optimum design can be better explained through an analogy. The search for the best optimum design can be likened to a blind man climbing to the top of a mountain. The blind man knows he is proceeding up the mountain by sensing the direction of ascent. However, the paths he takes may be limited by barriers or fences which will restrict the directions he can go. These fences represent constraints in the optimization problem. During the journey he may also encounter small crests and valleys. If the available paths lead the blind man up to one of these crests prior to reaching the mountain top, he will confronted with a situation where he will sense no further rate of ascent and he will stop his journey. So although he has made progress from his initial starting point, the man did not achieve his ultimate goal of climbing the mountain.

During optimization, the search for a true optimum may proceed along a path on which the objective function assumes such relative optimum values. If the optimizer can not be made to "look beyond" these relative peaks, then the optimization will cease - at a design which may be an improvement over the initial one but short of the true optimum. This problem may be overcome by starting the design with several different initial design vectors, X, until the same optimal design is repeated. Another alternative may be to increase the size of the finite difference so that the optimizer uses larger perturbations of X thus looking beyond any small increases in the objective function which could stand in the way of further design progress. This second



alternative will be specifically addressed during the discussion of the code development.

C. CONTROL PROGRAM FOR ENGINEERING SYNTHESIS (COPES)

The optimizer, CONMIN, was written in subroutine form. Vanderplaats [Ref. 13] has developed a main program to simplify the use of CONMIN and aid in the design optimization process.

The user must supply an analysis subroutine called ANALIZ, which consists of three segments: input, analysis and output. COPES acts as an interface between ANALIZ and the optimizer CONMIN. Based on a flag from COPES (ICALC=1,2,3) ANALIZ performs the proper function. Figure 2.1 offers a simplified illustration of the interrelationship between COPES, ANALIZ and CONMIN.

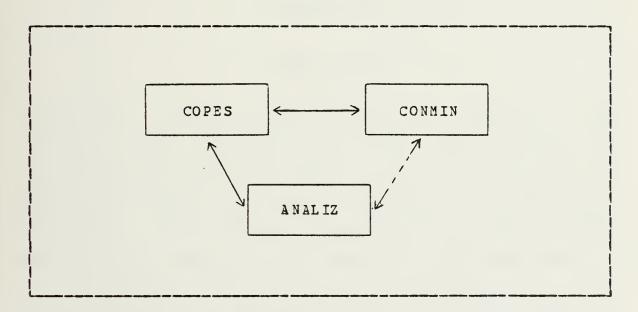


Figure 2.1 Flow Diagram For ANALIZ and COPES/CONMIN.

COPES currently provides four specific capabilities, two of which will be applied in this work:



- 1. Single analysis just as if COPES/CONMIN were not used.
- 2. Optimization minimization or maximization of a multivariable function with corresponding constraints.

COPES requires certain initial data from the user in order to coordinate the optimization process. Initial values for the design variables as well optimizer control parameters are utilized by CONMIN to conduct its numerical analysis. There are a few optimizer parameters which are particularly important to the treatment of condenser designs. One is the finite difference step used in gradient calculations. Another is the normalization factor used in COPES evaluation of a constraint function. COPES utilizes the following expressions in determining constraint function violations:

$$\frac{BU - CFV}{SCAL1} \leq 0.$$

$$\frac{CFV - BL}{SCAL2} \leq 0.$$

where SCAL1 and SCAL2 are the normalization factors. BU and BL are the upper and lower limits of the constraint, and CFV is the constraint function value. It is intuitive that the normalization factor can play an important role in determining the size of the active region about a given design constraint. Both finite difference and constraint normalization will be recalled later during the code development.

The power of COPES is that it has simplified the procedures involved in using a sophisticated program such as CONMIN. The user is therefore freed from the unwanted role of systems analyst and can concentrate on the design analysis.



III. CONDENSER DESIGN IMPROVEMENT PROGRAM (CONDIP)

A. BACKGROUND

In the late 1960's, engineers at the Oak Ridge National Laboratory developed a sophisticated computer code under contract to the Office of Saline Water. This code, called ORCON1, [Ref. 2] was generated to aid in the analysis and parametric study of large, generally circular condensers. Much of ORCON1 was dependent on Eissenberg's research work [Ref. 3] on the effects of condensate rain on the shell-side convective heat transfer coefficient. Johnson [Ref. 5] took ORCON1 and made a few minor modifications to determine tube side pressure losses and volumetric calculations. Nunn and Marto [Ref. 14] further incorporated the correlations proposed by Fujii [Ref. 4] to determine the effects of shearing forces exerted by high vapor velocities on the condensate film and resulting shell-side heat transfer coefficient.

It was at this point that the development of CONDIP was begun. CONDIP was dependent primarily on the principles detailed in ORCON1 but also incorporated subsequent developments to the basic program. CONDIP was written, however, in such a way as to be compatible with the optimizer, CONMIN.

condition analyzes a single or double pass, circular or semicircular condenser, with steam flowing radially inward on the shell-side of the tubes and variable salinity water flowing on the tube-side. An optional, rectangular air cooler bundle is provided-for as well as shell-side baffles. The circular bundle is normally divided into 30-degree sectors with symmetry about the central axis to reduce computational effort. Unless otherwise specified, tubes are



placed on a 60-degree equilateral, triangular pattern of concentric rows with the rows added from an inner void out to the outermost row. The void serves as a collection header for non-condensible gases prior to passage through an air cooler, if specified. As in ORCON1, CONDIP proceeds sector by sector, row by row through the condenser utilizing an average tube to represent the row segment, and calculates the following quantities in each sector:

- a) Steam pressure losses at the entrance of a sector.
- b) Total pressure of the steam/non-condensible gas mixture entering a row segment.
- c) Saturation pressure of steam entering a row segment.
- d) Saturation temperature of the steam entering a row segment.
- e) Steam flow entering the row segment.
- f) Velocity of the steam/non-condensible gas mixture at the minimum cross-section in the row segment.
- g) The fraction of non-condensible gas in the mixture by weight.
- h) The overall heat transfer coefficient for the average tube in the row segment.
- i) The steam-side condensing coefficient.
- j) The tube-side heat transfer coefficient.
- k) The shell-side film heat transfer coefficient composed of the non-condenible gas film and the condensate film.
- 1) The shell-side friction factor.
- m) The shell-side pressure loss as steam passes over the row segment.
- n) The shell-side Reynolds number based on the mass flow at the minimum cross-sectional area in the row segment.
- o) The heat transfer rate per square foot of condenser tube.
- p) The mass flow rate of steam/non-condensable gas mixture at the minmum cross-section in the row segment.



- q) The mass flow of condensate produced as steam passes ever the row segment.
- r) The cooling water temperature at the outlet end of the condenser.
- s) The coolant pressure loss on the tube-side.
- t) The average Reynolds number of the coolant through the tube.
- u) The heat transfer coefficient for the non-condensible gas film.
- v) The internal heat film heat transfer coefficient.
- w) The number of tubes per row segment.
- x) The cross-section area available for steam flow per row segment.
- y) The cumulative shell-side pressure drop.
- z) The LMTD based on inlet and outlet coolant temperatures and saturation temperature at each row segment.

In addition to the above parameters, the area-weighted overall heat transfer coefficients for the condenser, cooler and combined condenser are used to calculate the "back calculated" log mean temperature difference (LMTD). Steam exit-fraction, condenser volume, coolant pumping power and numerous other factors are calculated from the cumulative results of the row and sector analysis.

There are two significant contributions to the external film heat transfer coefficient which have a profound impact on the overall analysis. As mentioned earlier, Eissenberg [Ref. 3] corrected for condensate inundation effects on the external heat transfer coefficient with a series of emperical relations. He created a flooding factor F using the following relation:

$$F_n = .6F_d + (1 - .5647 F_d) n^S$$
 (eqn 3.1)



where \mathbf{F}_{d} is a constant indicating the effect of tube spacing and orientation on condensate side drainage. With closely packed tubes, significant side drainage can occur in low velocity steam flow. Condensate generated on tubes above may, due to surface tension effects, proceed laterally to adjacent tubes rather than down. Thus \mathbf{F}_{d} tends to approach 1.0 for closely packed, staggered tube bundles and zero for disperse bundle layouts. S is a constant ranging in value from:

If the condensate rain is acting under the influence of gravity alone S approaches 0.25. But the influence of any steam velocity present begins to alter the rate and direction of condensate flow and correspondingly decrease S. Thus S is a function of vapor velocity and directon, as well as bundle geometry.

The condensate film coefficient for the average tube in the n-th vertical rcw is then calculated from the uncorrected heat transfer coefficient, \mathbf{h}_0 , as follows:

$$h_n = [nF_n - (n-1)F_{n-1}]h_0$$
 (eqn 3.2)

It is obvious that the determination of a corrected heat transfer coefficient is highly dependent on the choice of S an F_d in equation 3.1. S and F_d are extremely subjective constants and there does not exist a current analytical expression to determine them. Yet the choice of these constants can have profound impact on the condenser design. In CONDIP, as in ORCCN1 [Ref. 2], the following, relatively conservative values for S and F_d were used:

$$S = 0.2$$
 $F_d = 0.5$



An additional important contribution to the external film coefficient is the effect of velocity shear forces on the condensate film. Fujii [Ref. 4] developed the following experimental correlations to correct the Nusselt number for the effect of velocity shear:

$$[Nu_{m}/Nu_{o}] = c_{1}Nu_{o}^{(4a-1)} Re_{L}^{(.5-2a)}$$
 (eqn 3.3)

where $\mathrm{Nu_m}$ is the mean Nusselt number. Re L is the two-phase Reynolds number (based on vapor velocity, tube outside drameter and kinematic viscosity of the condensate) and $\mathrm{Nu_0}$ is the standard Nusselt number for the zero shear case. The empirical constants $\mathbf{c_1}$ and a lie within the following ranges:

1.13
$$< c_1 < 1.24$$

0.196 $< a < 0.2$

depending on how tube thermal conditions are described. In CONDIP the values for c_1 and a were:

$$c_1 = 1.24$$
 $a = 0.2$

It should be noted that equation 3.3 is only valid in the range:

$$3.3 < (Re_1/Nu_0) < .28.$$

For smaller values of this parameter Fujii recommends the use of a slightly reduced value of the standard Nusselt number:

$$Nu_{m} = 0.96 Nu_{o}$$
 (eqn 3.4)

It is apparent that the vapor velocities commonly encountered in naval condensers can have an impact on the heat transfer coeffcient.



B. CODE DEVELOPMENT

Johnson [Ref. 5] attempted to couple OPCODE2 (his version of ORCON1) with the optimizer CONMIN, but with little success. There were several reasons for this.

ORCON1 uses iterative techniques to solve for such quantities as condensate rate, steam mass flow rate and steam pressure through the sector. If unrealistic values are encountered, such as negative pressures or steam flow, or if the final steam exit-fraction exceeds a predetermined value, ORCON1 stops the analysis, returns to the beginning of the program and changes certain initial input parameters. The analysis begins again and the process is repeated until a satisfactory design is achieved. Thus ORCON1 has a limited capability to make design decisions to obtain a feasible design.

CONMIN, as do most optimizers, requires complete control in determining all iterative design variable values. As explained earlier, it uses perturbation techniques to calculate gradient information for each design variable and active design constraint, which it then uses to determine search directions. A perturbation of the design variable by CONMIN requires a complete, once-through analysis. If ORCON1 is coupled with CONMIN then any adjustment by ORCON1 will yield false gradient information to the optimizer and hinder, if not completely prevent, CONMIN from arriving at the optimum design. During program development it became apparent that the two programs were working independently against each other and that in its present state ORCON1 was incompatible with CONMIN.

In the formulation of CONDIP, it was necessary to locate and neutralize all the places where such design decisions are made. By removing the ability for CONDIP to make any design decisions it became totally passive and dependent on CONMIN for design variable changes.



However, once this was accomplished, another problem area was discovered. In the ORCON1 code and subsequently in CONDIP there are numerous thermal process and properties calculations that use logarithmic functions and other mathematical relationships which could produce singularities if the variables in the arguments approach zero or are negative. For example, saturation steam temperature is calculated from steam pressure using a logarithmic relationship. If, during a design analysis, saturation pressure approaches a negative value, this represents a clear violation of physical realities and of the limits of that property. Yet a computer cannot make that distinction so it tries to calculate the corresponding saturation temperature which, because of the logarithmic relationship, would be undefined. As just explained, ORCON1 with its built-in decision capability simply starts over when this situation is encountered. But CCNDIP, being completely dependent on CONMIN for design decisions, does not have that capability. Remembering that CONMIN requires a complete once-through analysis in order to collect enough information to make a design decision, it was necessary to somehow bypass such mathematical instabilities in order to keep the program operating. Yet the analysis still had to yield reasonable results from the given design in order to obtain meaningful gradients. This prompted the formulation of mathematical relationships to create "penalty" constraints which, if properly written, would indicate to the optimizer that a function or thermal property has violated its physical limits. However, not only would penalty constraints have to be defined, but a "fix up" or "correction" of the violated property or variable would be required in order to allow the analysis to continue. A good physical understanding of the inter-relationship between condenser parameters and the thermal processes resulting from the condenser design is



necessary so that the "fix up" of the violated property would still yield fairly accurate condenser information on which CONMIN could base its search for the optimized design.

For example, a condenser is usually designed around a given steam load. If the condenser has too many tubes, is too long, or coolant flow is too great, then the condenser will be overdesigned. There will be dry tubes within the condenser as all the steam is condensed before steam flow reaches the inner void. In CONDIP this means that zero or negative steam flow will be encountered in the analysis.

If, on the other hand, tube surface area is too small, coolant flow is inadequate or the condenser tube spacing is too tight and is choking the steam flow, then one of two things will happen. Either a quantity of uncondensed steam will make it completely through the condenser, or steam pressure loss in the condenser will cause steam pressure to drop below zero. In addition, there are two reasons why all the steam might not condense. It could be simply due to insufficient heat transfer surface area or it could be because the saturation temperature of the steam has dropped below the coolant inlet temperature. If the latter situation occurs, then there is no driving force for heat to transfer from the steam to the coolant. There is only one way that this situation can occur: if steam saturation pressure drops below some value indirectly determined by the coolant inlet temperature. In any event this condenser is certainly underdesigned and not capable of supporting the required steam load.

As stated earlier, the purpose of optimization is to obtain the best, feasible design. Thus, an understanding of the relationship between the physical characteristics of a given design and the subsequent thermal performances will certainly help in defining the appropriate penalty constraints and their corresponding limits. It will also



aid in the determination of appropriate "corrections" when those limits are violated. It is important to note that with the introduction of these penalty constraints, the definition of a feasible design is revised. A feasible design is now defined as one in which thermal properties and functions are not allowed to violate their physical limits, as well as other design constraints, anywhere in the condenser.

In CONDIP there are three basic thermal properties which could create the above mentioned problems if they fall below a certain value. They are steam saturation pressure, steam flow and steam temperature. Because of the direct relationship between steam saturation pressure and temperature, it was possible to deal with them simultaneously. The solutions that were developed to overcome the effects of these thermal violations determined the extent which CONDIP would optimize.

1. Steam Flow Effects

One source of mathematical instabilities within the program is if steam flow over the tubes falls to zero or below. It is intuitively obvious that steam flow can not physically fall below zero and that in order to keep the program running the steam flow rate must be kept greater than zero. However, correcting for this alone would certainly alter the results of that particular condenser design, perhaps even imply a feasible design.

To indicate to the optimizer an infeasible design was actually encountered - one in which steam flow had dropped below zero - the penalty function WTST was created. Since the condenser analysis is performed sector by sector and assuming there are J sectors in the condenser, then there had to be arrangements for J penalty constraints. This prompted the creation of the array, WTST(J), representing constraint penalty functions for each sector. The absolute



magnitude of these functions were directly related to both the severity of the steam flow violation and the number of dry tubes remaining in the sector. WTST(J) ranged in value from negative infinity to zero, where a value of zero represented a condenser design in which no flow violations occurred. Thus, WTST(J) were constrained functions whose lower limit was zero. For example, during an analysis, if steam flow was determined to fall below zero, then WTST for that sector would be given some negative value. In subsequent designs, as the number of dry tubes approached zero and better designs were obtained, then the magnitude of the penalty function approached zero indicating no constraint violation.

Since penalty constraints are entirely contrived relationships with no real physical basis, it is desirable to minimize their number to avoid the possibility of sending innacurate signals to the optimizer.

To eliminate the need to use the WTST penalty functions as constraints, the values of WTST(J) were consolidated at the end of the sector analysis into the condenser steam exit-fraction constraint. Normally, steam exit-fraction ranges from zero to one and is simply equal to:

steam leaving the condenser steam entering the condenser

By incorporating the WTST violations into steam exitfraction, the exit-fraction was made a continuous function
ranging in value from negative infinity to one. The
negative steam exit-fraction represented a partially dry
condenser and its magnitude was in direct proportion to the
number of dry tubes. Thus instead of having to evaluate and
calculate gradients for J number of WTST(J) contraints, the
optimizer simply had to evaluate a previously defined and
now expanded exit-fraction constraint.



It should be stressed that making steam exitfraction continuous through zero was equally as important as eliminating the need for additional constraints. It can be reasonably assumed that for all practical condenser applications, exit steam fraction will always be limited to some positive number near zero. Here is where one applies the physical knowledge of the condenser and its relation to the thermal property of steam flow. As explained earlier, dry tubes represent an over-designed condenser. Thus the natural tendency is for the optimizer to alter those design variables so as to create a more compact condenser. As this occurs, steam exit-fraction will naturally increase. The upper active limit of that constraint will determine the optimum feasible design. While it is not necessary to have a lower limit for steam exit-fraction, it is very important for it to be a continuous smooth function especially in the region near the upper limit. It is therefore critical to properly define the penalty functions WTST(J) in a way so as to provide a smooth transition from the negative, artificial values of negative steam exit-fraction to the real, positive values.

Since the steam flow penalty functions will not be used as constraints, the analytical results will provide gradient information to the optimizer. However, once steam flow has been determined to fall below zero, steam flow for that first dry row of tubes and all subsequent rows must be fixed up with dummy values to allow the program analysis to continue. How that "fix-up" is accomplished will ultimately determine the search direction for the optimizer.

Physically, once steam flow has gone to zero, there should be no further latent heat transfer, no subsequent condensate production, and further pressure losses should be only due to the flow of non-condensible gases. It is necessary to make the computer generated analysis reflect as



closely as possible these physical realities. Since the optimizer no longer has the penalty functions to use in calculating a search direction, other constraint values obtained from the analysis will dictate the next search direction. Gradients will also be calculated using these results and the determination of the next search direction will incorporate these gradients as well. In the case of negative steam flow, steam flow and condensate production over dry rows were given nominal values which were as small as the computer analysis would tolerate. These extremely small values closely approximate zero steam flow and generate results which resemble physical reality as closely as possible.

The following example is provided to better illustrate the logic used in CONDIP to handle negative steam flow. CONDIP determines flow rate through each row in each sector. During a sector analysis, CONDIP calculates the condensate generated at a given row and subtracts that value from the steam flow entering that row to calculate the steam leaving. The exiting steam flow rate is then checked to determine whether steam flow has gone to zero. If it has not, then the analysis continues. If it has, then the following two events occur.

The penalty function, WTSF(J), is calculated for that sector and dummy values are inserted for steam flow and condensate rate at the row where the violation occurred. For the remainder of the analysis condensate generation and subsequent steam flow calculations are bypassed and the remaining rows in the sector are fixed up with dummy values for steam flow and condensate. The analysis continues utilizing these dummy values in all appropriate heat transfer and pressure calculations. At the conclusion of the sector analysis, the values of the penalty functions, WTST(J), of each sector are incorporated into the steam exit-fraction.



If the analysis revealed zero dry tubes then WTST(J) for all sectors would be zero and the steam exit-fraction would simply be calculated as:

steam leaving the condenser steam entering the condenser

If, however, dry tubes were encountered in the condenser analysis then WTST(J) of some or all the sectors would be negative and dependent in magnitude on the number of dry tubes in each of the J sectors, as well as the severity of the steam flow violation. Steam flow leaving any sector which has gone dry would be zero and steam exit-fraction would be evaluated as:

steam leaving any wet sectors steam entering the condenser

plus a weighted value of all the WTST(J) penalty function values. Using the relationships just described, it is apparent that steam exit-fraction: is negative if condenser tubes are dry; approaches zero as the design becomes feasible; and is greater than zero if there is steam leaving the condenser.

2. Steam Pressure and Temperature Effects

The other possible source of mathematical instability occurs when steam pressure falls below some preset limit. If pressure falls to zero, numerous mathematical singularities will be generated. Yet, before this situation can occur steam temperature will have already fallen below inlet coolant temperature causing singularities in the log mean temperature difference (LMTD) heat transfer calculation. Thus, the lower pressure limit which cannot be physically exceeded is not zero but the minimum saturation pressure established by the inlet coolant temperature. In CONDIP, this lower limit is given the variable name, PTLIM.



As steam flows through the condenser, pressure continually decreases due to friction losses and therefore, it is evaluated at each row in each sector. When the steam saturation pressure drops below PTLIM, indicating a physical violation of realistic limits, then the creation of a penalty function and a corresponding "fix up" of saturation pressure is required to allow the program to continue. The treatment of the problem was therefore analogous to the previous situation dealing with negative steam flow.

PTST(J) was the penalty function devised to indicate to the optimizer that the pressure limit, PTLIM, was violated in any of the J sectors. Values of these constraints ranged from negative infinity to zero, depending on the degree and location in the condenser of the violation. Since pressure is calculated on a row by row basis in each sector, the magnitudes of the pressure penalty functions were directly dependent not only on how much the calculated pressure dropped below PTLIM, but also on the number of rows remaining in the sector. Thus, as the condenser approached a feasible design the PTST(J) constaint values approached zero, indicating lessening violation of the minimum pressure.

As emphasized earlier, it is important to minimize the number of constraints, not only to avoid the possibility of sending confusing signals to the optimizer but also to reduce cost and improve program efficiency. This was accomplished here by inserting dummy values not only into the violated pressure variables but also associated thermal properties such as condensate generation, heat transfer coefficients and heat transfer rates for the row where the violation occurred and all subsequent rows in the sector. The dummy values were chosen such that realistic gradient information would be sent to the optimizer. The proper choice of "fix-up" values for these varaiables resulted in



the elimination of penalty functions as design constraints, and provided sufficient information to determine subsequent search directions.

It is necessary to understand the influence that steam pressure and temperature exert on the overall condenser analysis. With this knowledge it will be easier to predict the physical designs which could cause violations of the pressure limit. PTLIM is violated due to excessive steam flow pressure losses. As explained earlier, these large pressure losses would result from large steam velocities that are found in condensers which are too tightly designed. Thus, the particular condenser design is incapable of handling the required steam load, implying an infeasible design. Understanding this relationship will aid in choosing the appropriate "fix-up" values which will indicate to the optimizer that when the pressure limit is violated an infeasible condenser has been designed.

Physically, When steam temperature falls below coolant inlet temperature (PTLIM is violated) there is no heat transfer from the steam to the coolant and no additional steam is condensed. These physical realities must be reflected in the condenser analysis. Therefore, in subsequent rows, condensation and heat transfer rates were set equal to zero. Since there is no further condensation, the steam exit-fraction is equal to the steam flow at the point of violation divided by the total flow into the condenser. Thus PTLIM indirectly determines the exit steam fraction of the infeasible design. This relationship between exitfraction and the PTLIM violation is what makes the penalty constraints obsolete. If PTLIM is violated early in the steam's passage through the condenser, steam exit-fractions will be large, violating its upper constraint limit and thus reflecting an underdesigned condenser. As the condenser design improves, then exit-fractions will decrease.



Physically, this can only be accomplished if the condenser design "opens up", reducing pressure losses in the condenser. Consequently, as condenser designs become larger, steam exit-fractions decrease and the condenser is driven towards a feasible design.

The following example illustrates the logic employed by CONDIP to handle steam pressure and temperature violations within the analysis. Condenser inlet flow is divided by the number of sectors in the condenser. Condenser inlet saturation pressure is determined by the steam inlet temperature. Entrance pressure losses are calculated subtracted from the inlet pressure. The resulting pressure is checked against PTLIM and a violation at this point indicates a totally infeasible condenser in which no steam is condensed. Steam exit-fraction will thus be equal to one. If the saturation pressure is greater than PTLIM the analysis continues row by row through the sector. Pressure losses over each row are calculated and subtracted from the row inlet pressure to determine pressure into the next row of tubes. If this next-row steam pressure is determined to fall below PTLIM, then a thermal violation has occurred requiring "fix up". Subsequent rows are made to indicate zero condensate generation and zero heat transfer. Steam flow over the remaining rows is maintained at a constant value, which will subsequently be used to determine steam exit-fraction. Pressure variables over the remaining rows are given small positive values just large enough to allow the analysis to continue. Although all heat transfer and condensate calculations will be bypassed, the analysis must be allowed to continue so that pressure losses will continue to be calculated based on the steam flow at the point of violation. This is important since steam flow adjustments to the sectors are based on certain pressure comparisons between the sectors. The cumulative sum of all row pressure losses



in each of the sectors must be equal to within some tolerance. If they are not then steam flow into each of the sectors is altered so that the exit pressures from each sector converge to some common value. Thus, an accurate reflection of true pressure losses is important to this calculation.

The value of the steam exit-fraction is again determined to be the single constraint necessary to drive subsequent condenser designs to a feasible optimum configuration. The pressure penalty constraints proved to be superflous information, but the corresponding variable "fix-up" was critical in the determination of search direction.

C. LIMITATIONS

During the development of CONDIP, it became apparent that steam exit-fraction would become the key constraint during optimization of any objective function. A feasible design implies that steam exit-fraction is a small positive number perhaps somewhere between zero and 0.1 percent. explained earlier, violations of either steam flow or pressure physical limits resulted in penalty functions and variable "fix-up" which were later directly or indirectly incorporated in the calculation of steam exit fraction. Thus any feasible design, let alone the optimum one, centers on the limits placed upon this design constraint. Any number of design variable combinations will yield a feasible design, and each design variable affects steam exit-fraction differently. The intertwined, complex calculations used to ultimately determining exit-fraction are done by sector and row with each design variable repeatedly playing a factor. For example, the profound effect of both vapor shear condensate inundation on the shell-side heat transfer coefficient and consequently steam exit-fraction, is



indirectly determined by numerous design variables. However, their effects are impossible to predict. The cascading effect of the thousands of calculations performed during the course of a design analysis is to ultimately create a single, highly non-linear variable in the form of the steam exit-fraction, upon which design decisions will be made.

As more design variables were involved in the analysis, the cptimizer had difficulty determining their often conflicting effects on both the objective function and the steam exit-fraction. A small perturbation of each of the design variables independently would yield gradients indicating design improvement. But when these gradients were evaluated simultaneously to actually determine the direction of the subsequent design, their combined effect would actually indicate either no improvement of the objective function or a violation of the steam exit fraction design constraint. The end result would be that the optimization process would stall as no feasible search direction could be obtained. Larger perturbations to the design variables were required to properly evaluate their relative effects on the objective function and any active or violated constraints. This would enable the optimizer to overcome either small inconsistencies or discontinuities in the objective function and the constraint functions which would otherwise prevent the optimizer from reaching the optimum design. accomplished during data input by changing the normalized finite difference step from 0.01 to about 0.1. Increasing the finite difference is not without its drawbacks. As the optimum objective value is approached, the optimizer overlooks the subtle effects of small changes in the design variables because of the relatively large perturbations. Thus, depending on the initial design variables, optimizer will improve the design to some point near, but necessarily at, the optimum.



When a design becomes feasible, steam exit-fraction will always become an active constraint. But the stated goal is not in achieving a feasible design but in driving the design to a feasible optimum. However, this iterative process can not be accomplished at the expense of violating a constraint and it was here that further complication was introduced. The initial impetus in any optimization process is to first obtain a feasible design. However, once the very small steam exit-fractions are obtained that are necessary for a feasible design, the exit-fraction becomes extremely sensitive to any further design variable changes. Thus any effort to further improve the current design could easily cause exit fraction to increase. Even slight increases would be perceived as violations of the constraint limit and thus prevent further optimization from the first feasible design. There are two possible solutions to this problem. Either increase the upper limit on the exit-fraction constraint or redefine the constraint. COPES formulates the general constraint function in such a way as to allow the user to increase the active region about the constraint limit. This is accomplished here by increasing the normalization factor in the following expression for the exit-fraction constraint function:

$$\frac{BU - EXITFR}{SCALT} \leq 0.$$

where BU is the upper constraint bound. EXITFR is the exitfraction constraint value and SCAL1 is the normalization factor for this constraint. Increasing the normalization factor reduces the optimizer's sensitivity to constraint violations by enlarging the range of constraint values in which the constraint is active. This enlarges the region of feasibility and allows the optimizer more flexibilty in altering design variables by reducing the risk of violating



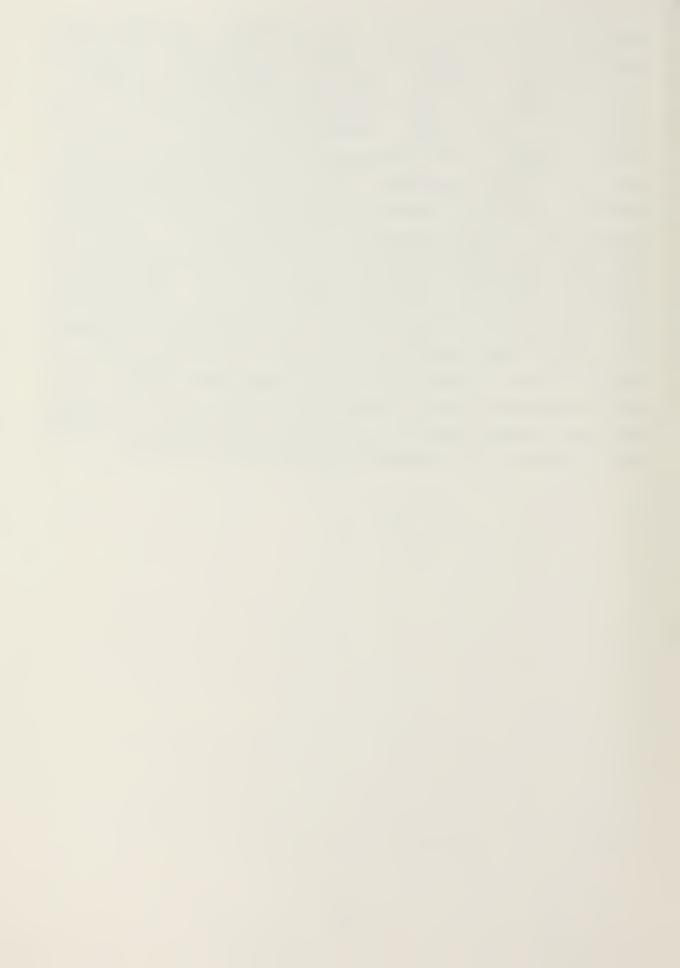
the constraint. The overall effect is that the design optimization can continue but at the expense of accurate constraint limits. The normalization factor used effectively for the exit-fraction in CONDIP analysis was approximately 0.1.

One of the stated objectives was to create a robust program which would consistently yield the single best optimum design independent of the initial design and not get hung up on a relative optimum. As it was explained earlier, although relative optimums represented design improvement, they also indicated the inability of the optimizer to locate the single best or true optimum design. However, the objective was achieved for only three design variables. When more than three design variables were used, the optimum designs became loosely dependent on the initial input, although not any predictable way. This is not to say that the condenser design did not optimize. By incorporating the finite difference and scaling normalization on exit-fraction as described above, final designs did yield objective function values which were continually within about ten percent of the true optimum regardless of the initial design. However, there was just no guarantee that the single, optimum design could be consistently obtained. In summary, the reasons why CONDIP did not consistently optimize to the single, true optimum were: the extreme non-linearity of the steam exit-fraction, the need for a large finite difference gradient, and the need for a normalization factor for the exit-fraction upper limit constraint.

while the optimum design solutions obtained from CONDIP may be sufficient, there are several ways to improve the results and increase the chances of obtaining the best possible design. The easiest way is to try several initial input values until the user is satisfied that the best solution has been obtained. The problem with this approach is



that it is both costly and time consuming. A second recommendation is to couple an extremely simplified version of the condenser analysis with the optimizer to obtain a educated guess as to what the optimum design should be. The results of this analysis could then be used as input for CONDIP. OPCODE1, which utilizes the HEI methods in its analysis, is a likely candidate. The advantage of this approach is that a quicker, cheaper analysis can be used to obtain a rough idea of the anticipated optimum design. CONDIP can then use these design results to obtain even better and more accurate solutions, faster. There is still no guarantee, however, that the true optimum will be solved. Perhaps with the development of more robust and versatile optimizers, ones which uses numerical techniques and methods that are better suited to this type of problem than CONMIN, precise solutions can be obtained. However, there is little more that can be done to simplify the analysis of the steam exit-fraction and subsequently linearize the problem.



IV. DESCRIPTION OF THE MAIN AND SUPPORTING SUBROUTINES

A. MAJOR SUBROUTINES

The following section contains a brief description of the major subroutines in CONDIP. The appropriate flow diagrams are also provided to better illustrate and complement the explanations. For further information concerning the various subroutines and functions see the CONDIP listing in Appendix C and ORCON1 [Ref. 2].

1. ANALIZ

This subroutine basically arranges CONDIP in a standardized form which is compatible with COPES/CONMIN. COPES uses a variable flag, ICALC, to coordinate the optimization process with ANALIZ. Utilizing this flag ANALIZ then calls the input, analysis and output portions of CONDIP as required. When COPES sets ICALC equal to one ANALIZ reads in all initial input. This is the only time any input can be entered. When ICALC equals two, COPES works with CONMIN to optimize the design. ANALIZ makes available the analysis portion of the program to be used repeatedly by CONMIN. When COPES sets ICALC equal to three, the optimization is complete and ANALIZ calls all applicable output subroutines. Figure 4.1 illustrates the flow process for ANALIZ.

2. INPUT

This subroutine enters all initial input of data by which the initial design is determined. The resultant design may be either feasible or infeasible, subject to the limitations previously discussed, so it is not critical what values are initially assigned to the design variables.



However, the initial input is screened to prevent the introduction of totally unrealistic values of variables into the program. For example, initial tube thickness, tube inner diameter, tube number and tube length are all checked to ensure that their values are greater than zero. If any of the screened initial inputs do not satisfy the minimum requirements, then the program exits prior to entry into the optimizer. The limits of the design variales and constraints will prevent similar situations from occurring during the analysis. Figure 4.2 presents the flow diagram for the INPUT subroutine.

3. <u>OUT3</u>

This subroutine simply prints all the initial values entered in the INPUT subroutine.

4. ORCON

This subroutine calculates the bundle geometry, flooding factors and such coolant flow parameters as pressure loss, flow rate, and pumping power. There are two options available to determine bundle geometry, each with certain advantages and disadvantages.

Option1: The number of rows is entered as a constant and the tube number is determined based on pitch, tube outer diameter, and row spacing. The advantage of this method to determine bundle geometry is that it allows the user to linearly vary pitch and/or tube inner diameter by row. The disadvantage is that tube number is a dependent variable. The optimzer is therefore limited in determining the optimum design by the specified number of rows.

Option 2: The number of tubes can be used as a design variable while the number of rows is determined by tube number, row spacing, pitch and tube outer diameter. There



is more flexibility in this method of condenser design but it is not possible to linearly vary tube pitch and inner diameter. The condenser bundle is generated from a specified inner void out, and all the appropriate condenser geometry is determined for one of the identical 30 degree sectors. Overall bundle volume is then calculated as is the ratio of tube hole area to tube sheet area.

Once the basic condenser geometry has been determined, the code then proceeds through an algorithm to calculate baffle location based on an input value specifying the number of baffles desired in the condenser bundle. After this has been completed, flooding factors are determined. That is, the number of tubes in a vertical row above the central tube in each row is calculated. This is done for each of the six sectors on one side of a circular bundle. Symmetry is assumed for the other side. These flooding factors are later used in calculations to determine the effect of condensate inundation on shell-side heat transfer coefficients.

Finally, ORCON calculates coolant mass flow, coclant velocity, header pressure difference and pumping power based on the type of coolant flow input received. The flow chart in Figure 4.3 is a simple illustration of the logic used in ORCON. From ORCON, the subroutine SECALC is called.

5. SECALC

This subroutine determines all the parameters of each of the sectors in the condenser by row. The first calculation made in SECALC is the determination of the cooler geometry, if there is one. Entrance pressure losses into the condenser bundle are calculated for each sector and saturation pressure is checked to ensure that it is greater that PTLIM. From this point, much of the remaining



subroutine is comprised of two do-loops with one nested inside the other. The outer loop cycles through however many sectors are in the condenser model. The inner loop cycles through the rows in the sectors. Pressure, temperature, mixture velocity, steam flow and condensate flow are calculated at each sector row. The subroutine HETTRN is called repeatedly to provide the necessary heat transfer information. Pressure and steam flow is checked continually at each row to ensure that neither falls below its predetermined lower limits. In the event that either situation occurs, the appropriate penalty function and fix-up procedure is implemented to enable the analysis to continue. As previously discussed, these values are chosen to reflect as accurately as possible real conditions which would occur when steam flow or pressure violate their physical limits.

Once all the sectors have been analyzed the cumulative steam-side pressure losses from each sector are compared. Steam pressure at the inner void must be uniform, therefore the sector pressure losses are required to be equal within some allowable tolerance. If they are not, then the distribution of inlet steam flow into each sector is altered to force the pressure losses to converge to a single value. Once steam flow to the sectors has been adjusted, the sector and row analysis in SECALC is repeated until the pressure losses within each sector approach a common value. After the pressure comparison has been satisfied, certain overall condenser parameters are calculated such as steam exit-fraction, bundle heat load, and steam-side pressure drops.

Finally, if a cooler is required, the subroutine COOLEX is called. Otherwise the condenser analysis is complete. The flow diagram for SECALC is presented in Figure 4.4.



6. HETTRN

This subroutine is called repeatedly in SECALC to solve for all shell and tube-side hear transfer properties for each row in each sector of the condenser. In particular, values for the overall heat transfer coefficient and log mean temperature difference are utilized by SECALC in computing condensate production and heat transfer rate at each row of tubes.

On entering this subroutine, a series of estimates for certain row variables are calulated. Based on an assumed initial value for the overall heat transfer coefficient, the exit coolant temperature and corresponding film temperature are calculated. Utilizing these temperatures, the LMTD, thermal resistances, individual heat transfer coefficients and numerous other heat transfer parameters are then calculated. Finally, another value for the overall heat transfer coefficient is determined based on the above-mentioned analysis, and this final value is subsequently compared to the initial value. If they are not in agreement, within a specified degree of tolerance, then the initial value for the overall heat transfer coefficient is updated and the entire process is repeated until the initial and final values converge. This iterative process is necessary as temperature dependent heat transfer coefficients, film temperature drops, and exit coolant temperatures are all being calculated simultaneously.

Note that it is in HETTRN that the concepts of vapor shear and condensate inundation are incorporated. Heat transfer coefficients are corrected for both effects based on the calculations presented earlier. Also note that since steam temperature is never allowed to drop below inlet coolant temperature in the calling subroutine SECALC, resultant LMTD calculations in HETTRN will not yield singularities.



Once all the heat transfer variables have been determined, control is returned to SECALC where the appropriate results are utilized and stored. The appropriate flow diagram for HETTRN presented in Figure 4.5.

7. COOLEX

This subroutine solves for all the necessary parameters required in the cooler analysis. The cooler is assumed to be of rectangular cross-secton with the height of the cooler not to exceed the difference between the condenser inner and outer radii. The values used for tube pitch and tube diameters in the cooler are the same as the innermost row of the condenser bundle.

Steam exits the condenser bundle, collects in the inner void and enters the bottom row of the cooler. The steam then proceeds vertically up through the cooler. The physical location of the cooler is not a prerequisite to the subsequent design, although it is expected that the cooler will be placed within the condenser bundle, thus the limit on cooler height.

The first calculation in COOLEX determines the steam velocity at minimum cross-secton in the first row of tubes, VLCMAX. VLCMAX is directly proportional to the amount of steam and non-condensible gas entering the cooler as well as the cooler geometry. Therefore the constrained limits for VLCMAX will play a major factor in the overall condenser design.

Subsequent row analysis is treated identically as in SECALC. However, all pertinent heat transfer data are calculated directly within COOLEX, making it independent of HETTRN. Steam pressure and steam flow are checked at each row to ensure that the appropriate limits are not violated and all thermodynamic parameters are calculated. At the conclusion of COOLEX, cooler performance variables such as



heat load, exit-fraction steam, steam pressure losses and overall heat transfer coefficients are calculated and control is returned to SECALC. The flow diagram for the COOLEX subroutine is illustrated in Figure 4.6.

8. <u>OUT2</u>

This subroutine prints the overall condenser bundle results including heat load, steam exit-fraction, overall heat transfer coefficient, overall condenser LMTD and bundle volume. Normally, OUT2 is called once after the initial design is analyzed and again after the optimum design has been determined. Final design variable values such as tube number, coolant flow, tube pitch, tube wall thickness and tube inner diameter are also printed.

9. OUT2C

This subroutine prints the cooler results as well as the combined cooler/condenser results. This subroutine is called from the subroutine OUT2 and is called only if a cooler is required and subsequently designed. Therefore, these results will always be printed in conjunction with OUT2 output.

10. <u>OUT3</u>

This subroutine prints a very detailed output of the condenser and cooler results by row and sector. Nearly all the thermodynamic and heat transfer properties are presented, thus providing a rather complete picture of conditions everywhere in the condenser. This is extremely helpful in determining, for example, where additional heat transfer enhancement would be most beneficial, or where baffles should be best located to reduce the effects of condensate inundation.



B. SUPPORTING SUBROUTINES

The following is a brief description of supporting functions and subroutines called frequently by the main subroutines.

- 1. <u>DFSVTY</u>: This subroutine returns the value of the mutual diffusivity of the steam and non-condensible gas present.
- 2. <u>XTR</u>: This function subroutine transforms the calculated data, received in the argument list, to the log values and performs a linear regression on two or more points using the model.
- 3. AMUFN: This function subroutine calculates the viscosity of the non-condensible gas in lbm/(ft-sec).
- 4. <u>BMUFN</u>: This function subroutine calculates the viscosity of a saline sclution in the range of 0-24 percent concentration and temperatures of 40-210 °F in lbm/(hr-ft).
- 5. <u>CPAFN</u>: This function subroutine returns a value for the heat capacity of the inert, non-condensible gas mixed in with the steam in units of Btu/(lbm-mol-of).
- 6. CPFN: This function subroutine calculates the specific heat of a saline sclution units of Btu/(lbm-oF).
- 7. <u>CPSFN</u>: This function subroutine calculates the heat capacity of steam in Btu/(lbm-mol-of).
- 8. <u>HFGFN</u>: This function subroutine returns a value for the latent heat of vaporization of water in Btu/lbm.
- 9. <u>PRSDRP</u>: This subroutine returns the shell-side pressure drop across a row of tubes in psia.
- 10. PSATEN: This function subroutine calculates saturation



pressure of steam as function of temperature. Pressure is returned in units of psia.

- 11. ROEFN: This function subroutine calculates the density of a saline solution of concentration range 0-24 percent and temperature range of 40-300 °F. Density is returned in units of lbm/(cu.ft.).
- 12. <u>SKBFN</u>: This function subroutine calculates the thermal conductivity of a saline solution of concentration range 0-24 percent and a temperature range of 40-300 °F. Thermal conductivity is in (Btu)/(hr-f-°F).
- 13. <u>TSATFN</u>: This function suroutine returns the value for steam temperature in OR given a pressure in psia.
- 14. <u>VGFN</u>: This function subroutine calculates the specific volume of steam as a function of temperature and pressure. It has units of (cu.ft.)/lbm.
- 15. <u>SWITCH</u>: This function subroutine reverses the order of a stored array.



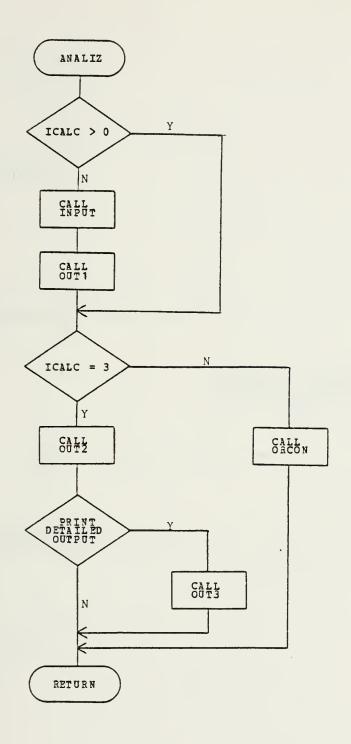


Figure 4.1 Plow Diagram for the ANALIZ Subroutine.



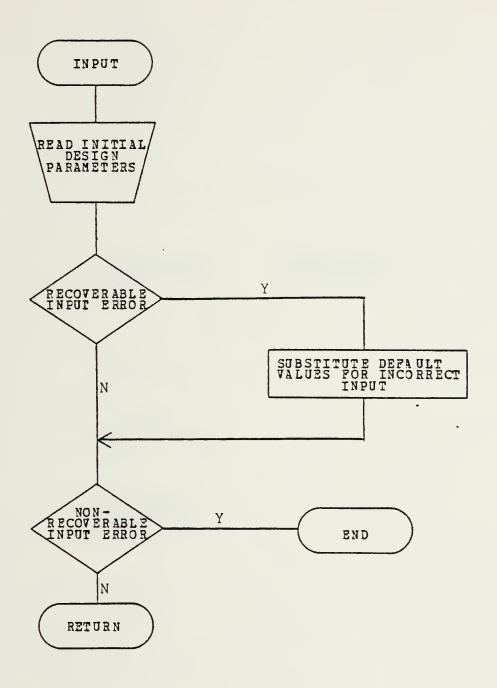


Figure 4.2 Flow Diagram for the INPUT Subroutine.



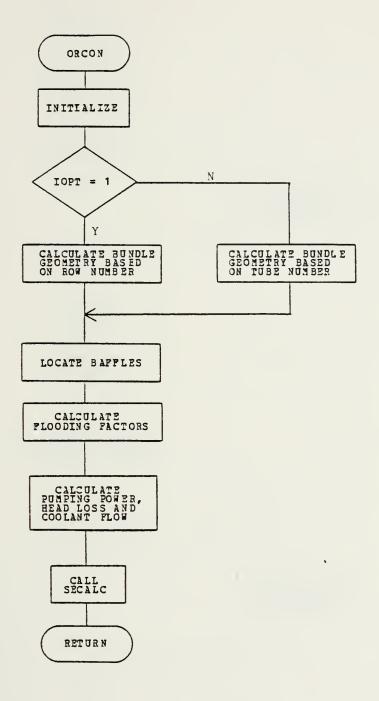


Figure 4.3 Flow Diagram for the ORCON Subroutine.



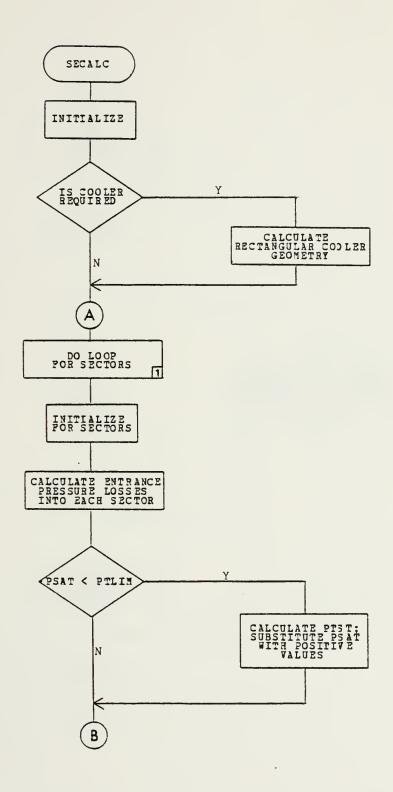
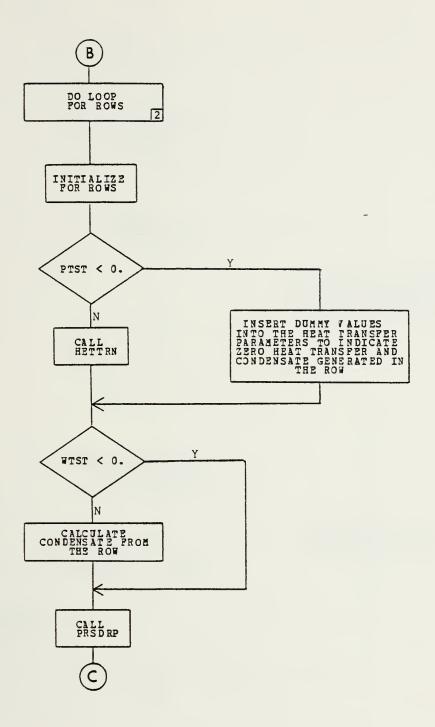


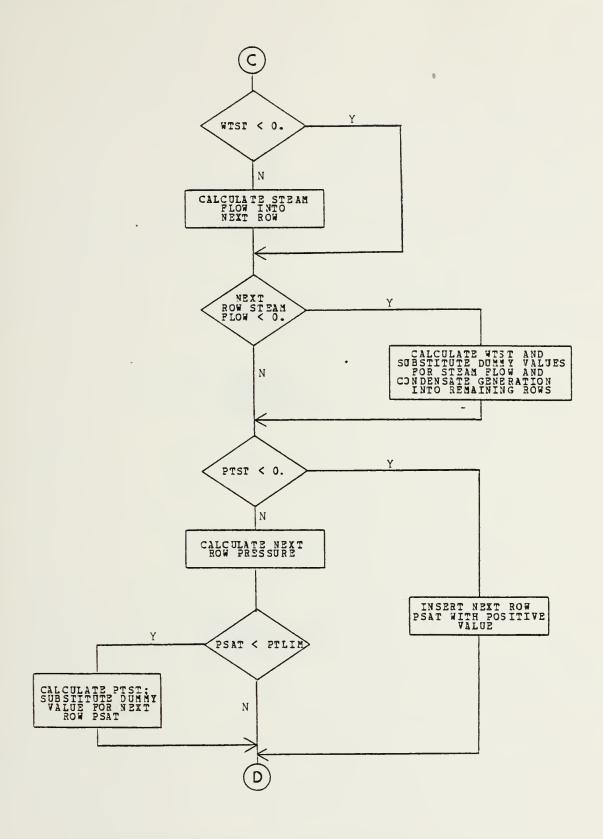
Figure 4.4 Plow Diagram for the SECALC Subroutine.





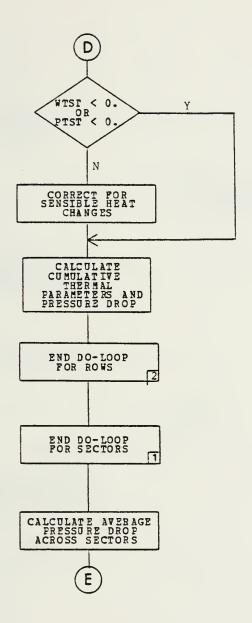
SECALC Flow Diagram (continued)





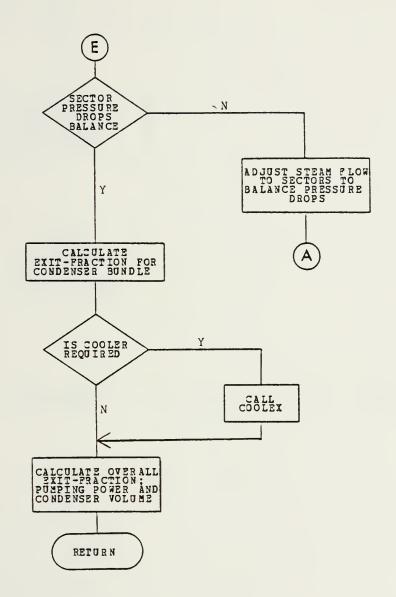
SECALC Flow Diagram (continued)





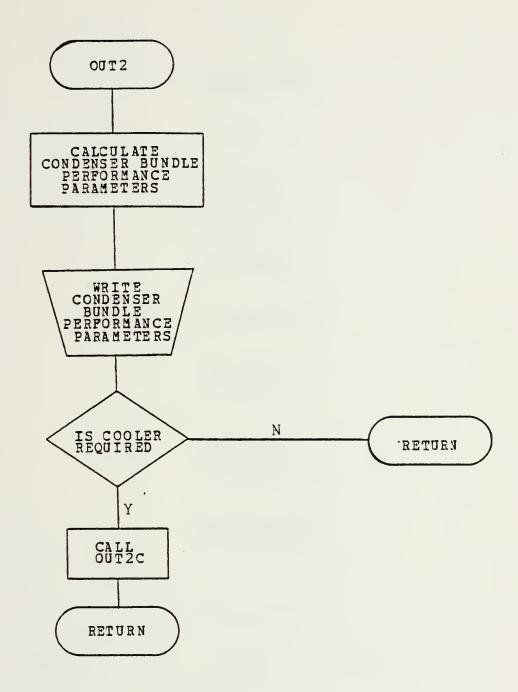
SECALC Flow Diagram (continued)





SECALC Flow Diagram (continued)





Flow Diagram for the OUT2 Subroutine



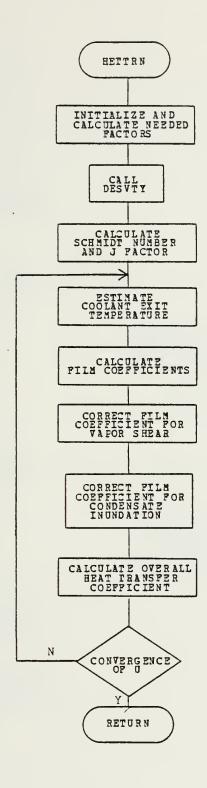


Figure 4.5 Flow Diagram for the HETTRN Subroutine.



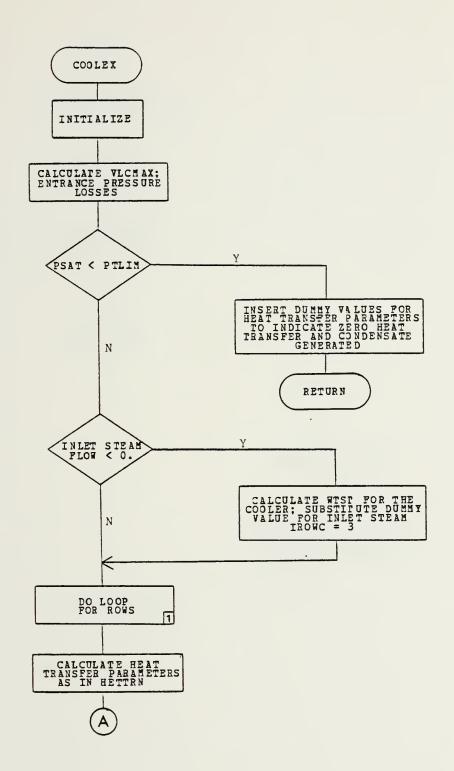
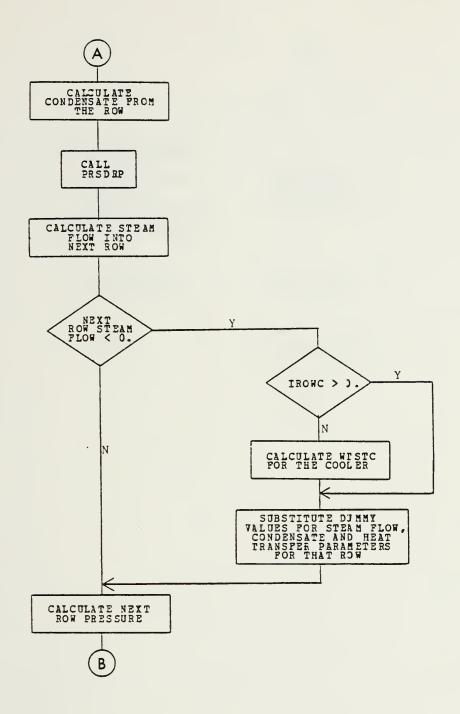


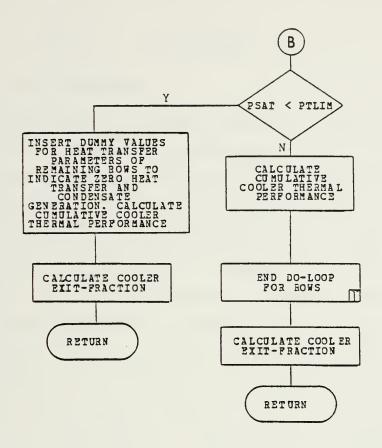
Figure 4.6 Flow Diagram for the COOLEX Subroutine.





COOLEX Flow Diagram (continued)





COOLEX Flow Diagram (continued)



V. RESULTS

A. CONDIP VERIFICATION

It was desirable to verify the single pass capability of without optimization) as a predictor of condenser performance by comparison with actual experimental However, complete and accurate data on condenser design and corresponding performance is not always readily available. Lynch [Ref. 18] encountered this same problem in attempting to verify ORCON1. However, he did manage to locate some actual experimental condenser data, obtained during a test conducted to determine the general performance of the DDG-37 class propulsion machinery [Ref. 16]. test took place at the Naval Boiler and Turbine Laboratory and was conducted primarily to determine the performance of the turbine and reduction gears. Some limited condenser data was taken as a by-product. The various measurements were obtained as described below:

- 1. Steam flow measurements were made by weighing the condensate.
- 2. Cooling water inlet and outlet temperatures were measured by thermometers installed in the inlet and discharge lines.
- 3. The heat load is calculated based on total steam flow into the condenser multiplied by the difference between inlet steam and condensate enthalpies.
- 4. Circulating water flow was determined from a heat balance around the condenser. The total heat load was divided by the circulating water heat capacity and temperature rise to obtain flow rate.



- 5. Condenser inlet pressure was determined by pressure instruments located above the condenser inlet flanges.
 - 6. Non-condensible gas flow was measured by a flowrator.
- 7. Pressure at the air ejector was measured directly. This pressure and the inlet pressure determined the pressure drop across the condenser.

It should be pointed out that this data were not recorded with the care that normally accompanies scientific data collection. Neither the instruments nor the techniques employed were particularly accurate. The possibility that this observed data are in error casts a cloud over the credibility of the corresponding condenser performance, which was calculated based on those values. However, for a lack of better alternatives, this data and the resulting condenser analysis will be used to determine the reliability of CONDIP.

The DDG-37 condenser geometric design variables obtained from the technical manual [Ref. 15] and input parameters corresponding to a full speed run are presented in Table I. An attempt was made to repeat the design using CONDIP. The results of CONDIP's proposed design as well as the experimental performance are presented for comparison in Table II. Percentages were calculated to quantify the differences between the actual and hypothetical performance.

Before elaborating on the results of this verification, some notable differences between the two designs must be clarified. First, a specific fouling factor was not determined at the time of the experiment and was therefore not provided. A somewhat realistic cleanliness factor of 87.5 percent (fouling factor of .0002) was utilized.

Second, in the DDG-37 condenser the rectangular cooler appears to be inserted directly into the condenser bundle.



However, in order to accomodate the cooler, the bundle must expand or distort. In addition, a void of some dimensions must be provided-for in the center of the bundle to collect any uncondensed steam and non-condensible gases. Diagrams in [Ref. 15] indicate that the DDG-37 condenser is indeed nearly elliptical in shape with bundle axes of 5.67 and 7.17 feet. Although it is apparent that a void does exist, exact dimensions can not be readily determined from the available information.

circular condenser bundle and a rectangular cooler, the height of which cannot exceed the difference between the outer and inner bundle radii. A circular void of predetermined size is provided-for when determining the condenser bundle geometry. Subsequent volume calculations are performed on the condenser bundle and cooler separately and the overall condenser volume is computed as simply their sum.

Although CONDIP does not exactly duplicate the geometric configuration of the DDG-37 condenser, it was possible to manipulate certain initial design variables in order to cause CONDIP to develop an approximately equivalent configuration. These variables were chosen because, for small changes in their values, there is a rather significant change in the bundle geometry with relatively small effects on the overall condenser performance. Since there are no specific dimensions provided for the inner void in the DDG-37 technical manual, it was picked to be one of the design variables to be adjusted. Row spacing was also adjusted because it satisfied the conditions described above. Through trial and error a combination of row spacing and inner void radius were determined, from which CONDIP yielded a geometric design similar to the DDG-37 condenser and that satisfied condenser requirements specified in



[Ref. 17]. In this particular case, the void diameter and row spacing were determined to be 1.1 feet and 1.35 inches respectively. This arrangement enabled the condenser model to closely approximate the tube sheet area ratio of the actual DDG-37 condenser. This manipulation, however, must be interpreted as another source of error and innacuracy when comparing CONDIP's condenser performance with the experimental results.

Lastly, condenser designs in [Ref. 15] reveal that three different tube patterns were employed in the DDG-37 condenser. In addition, two different values for tube pitch were used - a pitch of 1.4 in the condenser bundle and a pitch of 1.3 in the cooler. This situation cannot be duplicated in CONDIP. Therefore a constant pitch of 1.40 and a uniform tube pattern were utilized throughout the condenser.

The design approximations utilized in CONDIP to try to geometrically simulate the actual DDG-37 condenser introduce significant uncertainty into subsequent design comparisons. This, coupled with the fact that the data collected is also suspect, would imply that it is rather difficult to verify CONDIP's analysis with the information available. It should also be noted that CONDIP is sensitive to even small variance in either the data collected (i.e. steam inlet temperature) or the approximated design variables. However, despite the above-mentioned problems associated with equating CONDIP's condenser to the DDG-37 condenser, the experimental data obtained from the DDG-37 condenser still provide the best available base upon which to make a reasonable determination of CONDIP's capabilities and limitations.

In comparing the results in Table II, it is immediately clear that there is significant difference between certain condenser performance parameters predicted by CONDIP and the corresponding experimentally derived condenser performance. Already, much has been said about the numerous geometric



approximations used to model the DDG-37 condenser. But questionable data and geometric manipulations may not completely explain the 13 percent exit-fraction and general poor performance generated in CONDIP's analysis. Its values for the average overall heat transfer coefficient and heat rejected were significantly lower than the experimental results. One source of the problem lies in the actual heat transfer analysis performed in the code. Lynch [Ref. 18] graphically illustrated how sensitive this analysis is to the effects of condensate inundation. In particular, by making small changes - within the allowable ranges - in the constants used in Eissenberg's correlations for condensate rain, significant improvement could be realized in the overall heat transfer characteristics of the condenser. CONDIP's results, when compared to the experimental data support the argument that the values currently used in the inundation correlations are rather conservative in nature, and cause the overall analysis to yield a poor preformance for the given steam load and condenser design.

Therefore, in order to present CONDIP with a fair test to determine its credibility as a design predictor, some additional work must be first accomplished. A condenser geometrically identical to the general model created in CONDIP should be constructed with complete and accurate data acquisition systems to establish a thorough data base from which to compare. Also, more research should be performed on the effects of condensate inundation and velocity shear to obtain more precise correlations in determining their overall effects on the film heat transfer coefficients.

One last additional point should be mentioned. In comparing the steam-side pressure drops through the condenser, it was shown that CONDIP's pressure losses were nearly 72 percent larger the the actual physical measurements. However this radical difference is mainly due to the



high pressure losses experienced at the entrance of the cooler as a large volume of steam tried to force its way through the small available area. Therefore, the significance of this large disagreement in results is relatively minor and can be treated simply as a consequence of the more important heat transfer limitations in the comparison run.

Although the goal of verifying CONDIP as a design predictor has proven elusive, it was still possible to demonstrate its capabilities through comparison studies. Therefore, the remaining emphasis in this thesis is to show the ability of CONDIP (in combination with the optimizer COPES/CONMIN) to take an initial design with a given framework of constraints and design variables, and obtain better designs based on a desired objective function.

B. EXPLANATION OF THE CASE STUDIES

The following case studies were devised to best exercise the capabilities of CONDIP. They were made as realistic as possible so as to simulate the problem of condenser design and specification confronting the engineer during the early stages of power plant design. The condenser performance returned by CONDIP during the verification run and contained Table II will serve as a baseline for comparing the results of each case study. The baseline condenser performance is based on the design parameters from the DDG-37 condenser listed in Table I. It was stated earlier that CONDIP's optimization results are slightly sensitive to the initial design if more than three design variables are used. Since all the cases involve eight or more design variables it would be best, for the purposes of comparison, to start from the same initial design in all cases. Therefore, the initial design variables used for the verification run and contained in Table I will be utilized as the baseline



design. Although many of these initial design variables will be allowed to change during optimization, certain basic condenser requirements will not. They include: steam flow into the condenser, inlet steam saturation pressure and temperature, cooling water injection temperature, the fraction of non-condensible gases in the steam, the tube fouling factor, and the tube material. It should be noted that although there was an initial value for row spacing given in Table I, row spacing was not used as a design variable during any of the optimizations. Instead, the program used the default method of row spacing calculation available in the code where the rows are spaced such that a 60-degree equilateral triangle pattern of concentric rows is obtained. Row spacing is therefore dependent on tube pitch and tube outer diameter by the following relation:

$$RSPA = (SDDO * ODOI) * .866 (eqn 5.1)$$

where RSPA is row spacing, SDDO is tube pitch, and ODOI is tube outer diameter in inches.

There are a few key points to be kept in mind when comparing the results of the case studies with the baseline. First, the baseline design is an infeasible and inadequate design. Its performance indicates that it is not capable of supporting the required steam load by returning a steam exit-fraction in excess of 13 percent. So any gains in the objective function that were realized in the case studies is even more remarkable since it is a necessary condition that the optimum design be a feasible design, defined as having an exit-fraction not greater than 1 percent. Second, the percent change referred to when analyzing the results is calculated based on the baseline design. Thus the baseline serves as a uniform frame of reference. Next, it should be noted that because of the large number of design variables



and constraints, intuition on how an optimized result will turn out is not always applicable. Finally, it will be easier to understand the effects of the various design parameters by keeping in mind the following, very basic, heat transfer correlation:

$$Q = U * A * LMTD (eqn 5.2)$$

where Q is the rate of heat released as the steam condenses; U is the overall heat transfer coefficient; A is the heat transfer surface area; and LMTD can be interpreted as the thermal driving force between the steam and the coolant. Q is directly dependent on steam flow and pressure into the condenser, the percentage of that steam that is condensed, and any subcooling of the condensate. For a given steam load and a very small exit-fraction Q is nearly constant as the optimized results in all the case studies will indicate.

1. Constraint Framework for CONDIP

In order to simulate an actual trade-off study, the constraints and their respective limits were kept constant for all the case studies. The condenser was to be designed with a maximum bundle diameter of ten feet, a maximum and minimum tube outer diameter of 0.625 and 1.25 inches respectively, a steam exit-fraction of not more than 1 percent, a maximum cooler inlet velocity (VLCMAX) of 200 feet/second, and a ratio of tube sheet hole area to total tube sheet area of less than 0.30.

The constraint on bundle diameter was chosen somewhat arbitrarily. It seems unlikely that this limit would be realistically exceeded, although certainly space requirements would dictate the exact configuration. Tube cuter diameter is dependent on the values for tube wall thickness and tube inner diameter. Thus the limits imposed on tube



outer diameter represent realistic restrictions on the possible cobinations of inner diameter and wall thickness. These restrictions are based loosely on anticipated tube structural and strength requirements and correspond to values of normally available tubes [Ref. 19]. The maximum limit of 200 feet/second for VLCMAX was also a somewhat arbitrary but realistic limit. It is assumed that steam velocities often exceed that value in the condenser bundle.

It is recalled that steam exit-fraction will play a significant role in the determination of the final optimum design. The baseline exit-fraction of 13 percent predicted by CONDIP for the DDG-37 condenser is unsatisfactory. Therefore a more reasonable upper limit of 1 percent was placed on this constraint. Although CONDIP will return a much more conservative design if 1 percent vice 13 percent is used as the upper limit, the subsequent design will be much more credible.

Finally, the amount of tube sheet material that can be removed by drilling for the installation of condenser tubes is specified at 24 percent of the total tube sheet area in [Ref. 17]. This area ratio limit represents a structural limit imposed to ensure that the tube sheets do not fail due to heat and pressure stresses in the condenser. However, CONDIP does not take into account the space between the condenser and tube shell normally used in area ratio calculations as blank tube sheet area. For this reason and to allow more flexibility in the design analysis, the constraint limit was set at 30 percent.

In summary, the general design constraints and the associated upper and lower bounds were:

0.625 ≤ tube outer diameter (inch) ≤ 1.25
1.0 ≤ bundle diameter (feet) ≤ 10.0
steam exit-fraction (%) ≤ 1.0
VLCMAX (ft/sec) ≤ 200.0
area ratio ≤ .30



These design constraints and associated bounds were used in all the case studies except where specifically modified.

2. Design Variable Framework for CONDIP

At least eight design variables were used in all the case studies. They include tube inner diameter, tube wall thickness, tube pitch, the number of tubes in the condenser, tube length, the inner void radius, the percent of the tubes in the cooler, and cooling water velocity. Side constraints were placed on all of these variables to correspond to either realistic physical limits or available standardized materials.

Tube wall thickness was not allowed to fall below 0.022 inches (BWG 24) or exceed .109 inches (12 BWG), sizes normally available commercially. Tube inner diameter was restricted to values between .407 and 1.206 inches so as to yield tube outer diameters within the limits specified earlier.

Tube pitch is defined as the ratio of the center to center spacing between adjacent tubes in a row to the tube outer diameter. Tube pitch is an accurate measure of how closely packed the tube bundle is. Generally accepted values for pitch lie in the range of 1.3 to 1.7. However, to provide more latitude in the design process this design variable was allowed to vary in the range between 1.1 to 2.0.

There is no guidance available as to the allowable range for tube length in the condenser. Since the lower limit was not expected to be crucial, it was set randomly at 1.0 feet. The upper limit of 25.0 feet was a realistic limit considering the size of the tube diameter being worked with. Inner void radius and the percent of tubes in the cooler were chosen to be design variables simply to enhance the



flexibility of the code in designing the condenser model. The bounds for both variables were entirely arbitrary with only common sense as the determining factor. The upper and lower limits on the percent of tubes in the cooler was established as 10.0 and 2.0 percent respectively. The upper and lower bounds on the inner void radius was set at 1.0 and 0.1 feet.

Cooling water velocity generally ranges from three to nine feet per second in value for all common tube materials, except titanium which has an upper limit of 15 feet per second. Exceeding these upper limits risks excessive tube erosion and material damage. Finally the number of tubes was permitted to vary between 1000 and 8000 tubes for the purpose of improving design flexibilty. It is extremely unlikely that, for most propulsion applications, tube number would fall below 1000. The upper limit was simply chosen as a realistic cutoff point in terms of complexity, cost and maintainability.

In summary, the general design variables and the associated side-constraints were:

 $0.407 \le \text{tube inner diameter (inches)} \le 1.206$

 $0.022 \le \text{tube thickness (inches)} \le 0.109$

2.0 ≤ percent of tubes in cooler ≤ 10.0

 $0.10 \le inner void radius (feet) \le 1.0$

3.0 \leq coolant velocity (ft/sec) \leq 9.0

1.0 \leq tube length (feet) \leq 25.0

 $1000 \le \text{tube number} \le 8000$

1.1 \leq tube pitch \leq 2.0

As in the case of design constraints, these design variables and their respective limits were used consistently in all the case studies unless otherwise specified.



C. CASE STUDIES USING CONDIP

1. Case One

The objective of this case was to minimize condenser volume. The final results of the optimization along with the initial parameters is listed in Table III.

These results show a 16 percent decrease in condenser volume with a corresponding 24 percent increase in pumping power. The source of the improvement can be understood by noting the following:

- 1) Tube wall thickness was reduced from 0.049 to 0.022, the minimum side-constraint, thus allowing tube inner diameter to increase while maintaining a minimum tube outer diameter.
- 2) The number of tubes shrank slightly as did tube length, resulting in a smaller heat transfer surface area.
- 3) Tube pitch increased markedly, causing a reduction in steam pressure losses which then ensured that high values for steam saturation pressure and temperature would be maintained throughout the condenser. The large pitch also reduced steam velocities, allowing the cooler inlet velocity limit to be satisfied. Row spacing decreased from the initial value of 1.35 inches, thus decreasing condenser volume.
- 4) Cocling water velocity increased to the maximum allowable value of 9 ft/sec which correspondingly resulted in larger head losses and coolant flow, causing overall pumping power to increase.

As cooling water velocity increased and tube wall thickness decreased, then their respective thermal resistances were diminished. The cumulative effect was to improve the overall heat transfer coefficient. LMTD rose primarily as a result of the higher steam temperatures throughout the condenser. It is apparent by looking at equation 5.2 that increasing the driving forces for heat



transfer, such as the overall heat transfer coefficient and LMTD, allows the heat transfer surface area to decrease. This resulted in a similar reduction in condenser volume.

The constraint limits that prevented further design improvement were the upper bound on the cooling water velocity, the upper limit on the tube sheet area and the upper limit on VLCMAX.

2. Case Iwo

The objective of this case was to minimize the pumping power required to overcome the tube-side head losses and drive the cooling water through the condenser tubes. The final results of the optimization are presented along with the initial design in Table IV.

The results indicate a dramatic 90 percent reduction in required pumping power with an equally large 120 percent increase in condenser volume. The major factors involved in the design improvement along with their relative effects are briefly explained below:

- 1) Tube inner diameter increased 27 percent while tube thickness remained relatively unchanged. Thus tube cuter diameter was caused to increase.
- 2) The number of tubes in the condenser rose significantly, along with tube length. This, coupled with the enlarged tube outer diameter resulted in nearly doubling the heat transfer area.
- 3) Tube pitch icreased 29 percent, which allowed steam saturation pressure and temperature to be maintained at consistently large values in the condenser. This had a benefiting effect on the associated LMTD calculation. The large pitch also helped satisfy the steam velocity limit into the cooler. The tube spacing decreased from the initial value of 1.35, but by a smaller amount than the previous case because of the large values for tube pitch and outer diameter.



4) Cooling water velocity dropped to the minimum allowable limit of 3 ft/sec. This had the effect of reducing tube-side head losses and coolant flow through the condenser. Consequently, pumping power was drastically reduced.

The combined effect of all these changes can again be put in perspective by looking at equation 5.2. For the given steam and corresponding heat load, the heat transfer area increased drastically, allowing both LMTD and the overall heat transfer coefficient to decrease. A smaller overall heat transfer implies a smaller convective tube-side contribution which in turn permits coolant velocity to reduce to its lowest allowable value. The LMTD decrease is explained by the fact that cooling water was spending more time in the tubes, thus causing the average cooling water temperature to rise. However, the subsequent reduction in LMTD was minimized by the fact that a high steam temperature was maintained in the condenser.

There were no active constraints in this design outside of cooling water velocity which prevented further design improvement. However, the penalty paid in terms of a huge condenser volume, appears prohibitive.

3. Case Three

The objective of this case was to minimize pumping power while holding condenser volume constant at the initial value of 432 cubic feet. This was a particularly interesting test case as the results in Table V bear out. The required pumping power was reduced by nearly 38 percent with no change in volume. The effects of the design changes which resulted in the design improvement are presented below:

1) Tube inner diameter increased noticeably. However, the effects of this increase on tube outer diameter was minimized by a large drop in tube wall thickness. Thus, tube outer diameter remained relatively unchanged.



- 2) The number of tubes experienced a minor reduction, while tube length increased. The overall effect was to increase heat transfer surface area.
- 3) Tube pitch again rose by nearly 25 percent, causing steam saturation temperature and pressure to maintain a nearly constant value throughout the condenser. This had a beneficial effect on the LMTD between the steam and the cooling water. The larger pitch also had the additional effect of reducing steam velocity thus allowing the subsequent design to satisfy the upper limit on steam velocity into the cooler (VLCMAX). The combination of tube pitch and tube cuter diameter resulted in a reduction in row spacing from the initial value of 1.35
- 4) Cooling water velocity decreased by about 21 percent. This effect was manifested in subsequent pressure head, coolant flow and pumping power calculations.

Looking at equation 5.2 we see the same general pattern emerging as in Case 3, but with more subtlety in the changes. Heat transfer increased, but not at the expense of volume. Cooling water velocity was allowed to decrease while the overall heat transfer coefficient actually rose. One explanation is that as the tube wall got thinner its thermal resistance got smaller which more than offset the loss of convective heat transfer contribution from the coolant. The LMTD dropped slightly due to the higher average coolant temperature of the coolant in the tubes.

The constraints which became active and prevented further improvement in the design include tube sheet area ratio as well as tube wall thickness. However, tube wall thickness was particular crucial because of its related effect on heat transfer.



4. Case Four

The objective of this case was to minimize condenser volume while holding pumping power constant at the initial value of 55.7 horsepower. The results of this case can be found in Table VI. Chosen to contrast the results in Case 4, the relative improvement in this design objective was not nearly so impressive. Condenser volume shrank by only 12 percent. An explanation of the causes and effects is provided below:

- 1) Tube inner diameter increased, with a corresponding decrease in tube wall thickness to yield the minimum allowable tube outer diameter.
- 2) The number of tubes decreased noticeably, tending minimize bundle volume. Note, there was only slight increase in tube length. The overall effect was to similarly reduce heat transfer surface area as condenser volume decreased.
- 3) Tube pitch again increased significantly, having the same effects on steam pressure, temperature and steam velocity into the cocler as discussed earlier. A large tube pitch benefits the LMTD between the steam and the cooling water. Row spacing was again a factor in reducing condenser volume as before.
- 4) Cooling velocity decreased slightly as did head loss. But overall coolant flow increased due to an increase in tube inner diameter. The net effect was to maintain pumping power.

Again, referring to equation 5.2, it is clear that the slight decrease in heat transfer area was offset by the slight rise in LMTD resulting from higher condenser steam temperatures. The significant improvement in overall heat transfer coefficient, therefore, is what makes the heat balance work. The large decrease in tube wall thickness and corresponding reduction in thermal resistance contributed heavily to this improvement.



There were several contraint limits which prevented any additional objective optimization. They include the minimum tube wall thickness, tube sheet area ratio and steam velocity entering the cooler.

5. Case Five

The objective of this case was to minimize condenser volume while exercising CONDIP's capabilities to linearly vary tube pitch and tube inner diameter by row. Thirty-five rows were used, which was the identical number as the initial design. Tube pitch and tube inner diameter of both the outermost and innermost rows served as design variables in this case. However - because tube number is now a dependent variable based on the number of rows, tube pitch, and tube diameter - it could not be used as a design variable. The optimized results of this analysis along with the initial design are presented in Table VII.

The results of this test case indicate a condenser volume which is 20 percent smaller than the initial design as compared to a 16 percent decrease in Case 1. The basic reasons and explanations as to why volume was able to be reduced remain fundamentally the same as in Case 1. Attention will therefore be focussed on the effects of linearly varying pitch and tube diameter. Final tube pitch ranged in value from 1.75 in the inner row to 1.44 in the outer row. Similarly, tube inner diameter ranged from .729 in the inner row to .583 in the outer row.

It is believed that smaller pitch and inner diameter were used in the outer row because of the higher available steam saturation pressure and temperature. The resulting higher steam velocities enhanced the beneficial effects of vapor shear on the external heat transfer coefficient thereby improving heat transfer on the outer rows. As steam pressure decreased, then tube pitch and tube inner diameter



increased to compensate and extract all the available heat from the steam. Consequently, steam velocity decreased and was able to satisfy to the limit imposed on the cooler entrance velocity. The end result is a condenser geometry that makes complete use of the available resources and conforms to the geometry to take advantage of the thermal conditions in the condenser. The big limitation with this approach is that the number of rows is held constant. Thus the subsequent condenser is designed around that value and the subsequent optimum design is a function of the number of rows specified.



TABLE I
Input Design Data

PARAMETER	VALUE
Total number of tubes	5230
Tube length (feet)	10.3
Tube inner diameter (inches)	0.527
Tube wall thickness (inches)	0.049
Tube outer diameter (inches)	0.625
Tube mat'l thermal conductivity btu/(ft-hr-0F)	26.0
Tube pitch	1. 38
Percent of tubes in the cooler	7.0
Steam inlet flow (lbm/hr)	161,961
Fraction of non-condensible gas (ppm)	37.1
Steam inlet pressure (psia)	1. 294
Steam inlet temperature (°P)	110.52
Coolant inlet velocity (ft/sec)	8.473
Coolant inlet temperature (°F)	75.66
Fouling factor	.0002
Inner void diameter (feet)	1. 1
Row spacing (inches)	1.35



TABLE II
CONDIP Verification Results

FARAMETER	EXPERIMENT RESULTS	CONDIP RESULTS	CHANGE (%)
Heat transfer area (sq.ft.)	8805	88 14	+0.10
Overall heat transfer coefficient btu/(hr-sq.ft0F)	635.2	547.9	- 9.5
Log mean temperature difference (°F)	28.24	28.62	+1.3
Coolant temperature rise (°F)	10.61	9.90	-6.7
Coolant mass flow rate (107 lbm/hr)	1.503	1.540	+2.5
Condenser volume (cu.ft.)		432.3	
Bundle diameter (ft)	5.7 7.2	7.17	
Shell-side pressure drop (psia)	0.751	1.29	+71.8
Steam exit-fraction (% of input)		13.3	
Heat rejected (10° btu/hr)	1.595	1.451	-9.03
Area ratio	0.291	0.266	₹8.59



TABLE III
Volume Minimization

PARAMETER	BASELINE RESULTS	OPTIMIZED RESULTS	CHANGE (%)
Total number of tubes	5230	5117	-2.2
% of tubes in cocler	7.0	7.01	+0.1
Tube length (ft)	10.3	9.92	-3.7
Tube inner diam. (in)	.527	.582	+10.4
Tube wall thick. (in)	-049	.022	-55.1
Tube outer diam. (in)	.625	.626	+0.2
Tube pitch	1.4	1.73	+23.6
Void diameter (ft)	1.10	1.34	+21.8
Bundle diameter (ft)	7.17	6.59	-8.1
Condenser volume (cu.ft.)	432.3	362.2	-16.2
Area ratio	0.266	0.300	+12.7
Coolant inlet vel. (ft./sec)	8 .47 3	9.00	+6.2
Coolant mass flow rate (107 lbm/hr)	1.540	1.955	+26.9
Head loss (ft H2O)	7.35	7.18	-2.3
Pumping power (hp)	55.69	68.99	+23.9
Coolant temperature rise (°F)	9.90	9.00	- 9.1
Log mean temperature difference (°F)	28.62	29.13	+1.8
<pre>Heat transfer area (sq.ft.)</pre>	8814.	8320.	-5.6
Average overall heat transfer coefficient btu/(hr-sq.ft0F)	574.9	689.9	+20.0
Steam exit-fraction (% of input)	13.3	0.0	-100.
Heat rejected (10° btu/hr)	1.451	1.672	+15.2



TABLE IV
Power Minimization

PARAMET ER	BASELINE RESULTS	OPTIMIZED RESULTS	CHANGE (%)
Tctal number of tubes	5230	6393	+22.2
% of tubes in cocler	7.0	7.4	+5.7
Tube length (ft)	10.3	13.34	+29.5
Tube inner diam. (in)	.527	.667	+26.6
Tube wall thick. (in)	.049	.0455	-7.1
Tube outer diam. (in)	.625	.758	+21.3
Tube pitch	1.4	1.812	+29.4
Void diameter (ft)	1.10	1.13	+2.7
Bundle diameter (ft)	7.17	9.25	+29.0
Condenser volume (cu.ft.)	432.3	964.4	+123.1
Area ratio	0.266	0.277	+4.1
Ccolant inlet vel. (ft./sec)	8.473	3.00	-64.6
Coolant mass flow rate (107 lbm/hr)	1.540	1.067	-30.7
Head loss (ft H2O)	7.35	1.11	-84.9
Pumping power (hp)	55.69	5.84	-89.5
Coolant temperature rise (°F)	9.90	16.3	+64.6
Log mean temperature difference (°F)	28.62	24.82	-13.3
Heat transfer area (sq.ft.)	8814.	16,921	+92.0
Average overall heat transfer coefficient btu/(hr-sq.ft0F)	574.9	393.8	-31.5
Steam exit-fraction (% of input)	13.3	1.0	-92.5
Heat rejected (10° btu/hr)	1.451	1.654	+14.0
(10° btú/hr)			



TABLE V

Power Minimization With Volume Constant

PARAMET ER	BASELINE RESULTS	OPTIMIZED RESULTS	CHANGE (%)
Tctal number of tubes	5230	5062	+3.2
% of tubes in cocler	7.0	6.7	-4.3
Tube length (ft)	10.3	11.39	+10.6
Tube inner diam. (in)	.527	. 594	+12.7
Tube wall thick. (in)	.049	.022	-55.1
Tube outer diam. (in)	. 625	.638	+2.1
Tube pitch	1.4	1.753	+25.2
Void diameter (ft)	1.10	1.14	+3.6
Bundle diameter (ft)	7.17	6.73	-6.1
Condenser volume (cu.ft.)	432.3	431.6	-0.2
Area ratio	0.266	0.297	+11.7
Coolant inlet vel. (ft./sec)	8.473	6.74	-20.5
Coolant mass flow rate (107 lbm/hr)	1.540	1.508	-2.1
Head loss (ft H2O)	7.35	4.69	-36.2
Pumping power (hp)	55.69	34.77	-37.6
Coolant temperature rise (°F)	9.90	11.6	+17.2
Log mean temperature difference (°F)	28.62	27.63	-3.5
Heat transfer area (sq.ft.)	8814.	9631.4	+9.3
Average overall heat transfer coefficient btu/(hr-sq.ft0F)	574.9	627.3	+9.1
Steam exit-fraction (% of input)	13.3	0.0	-100.
Heat rejected (10° btu/hr)	1.451	1.669	+15.0



TABLE VI
Volume Minimization With Power Constant

PARAMET ER	BASELINE RESULTS	OPTIMIZED RESULTS	CHANGE (%)
Total number of tubes	5230	4867	-6.9
% of tubes in cocler	7.0	7.2	+2.9
Tube length (ft)	10.3	10.92	+6.0
Tube inner diam. (in)	.527	. 581	+10.2
Tube wall thick. (in)	.049	.022	-55.1
Tube outer diam. (in)	.625	.625	0.0
Tube pitch	1.4	1.739	+24.2
Void diameter (ft)	1.10	1.20	+9.1
Bundle diameter (ft)	7.17	6.41	-10.6
Condenser volume (cu.ft.)	432.3	378.4	-12.4
Area ratio	0.266	0.299	+12.4
Coolant inlet vel. (ft./sec)	8.473	8.27	-2.4
Coolant mass flow rate (107 lbm/hr)	1.540	1.701	+10.5
Head loss (ft H2O)	7.35	6.67	-9.3
Pumping power (hp)	55.69	55.80	+0.2
Coolant temperature rise (°F)	9.90	10.34	+4.4
Log mean temperature difference (°F)	28.62	28.38	-0.8
<pre>Heat transfer area (sq.ft.)</pre>	8814.	8697.	-1.3
Average overall heat transfer coefficient btu/(hr-sq.ft0F)	574.9	677.4	+17.5
Steam exit-fraction (% of input)	13.3	0.0	-100.
Heat rejected (10° btu/hr)	1.451	1.672	+15.2

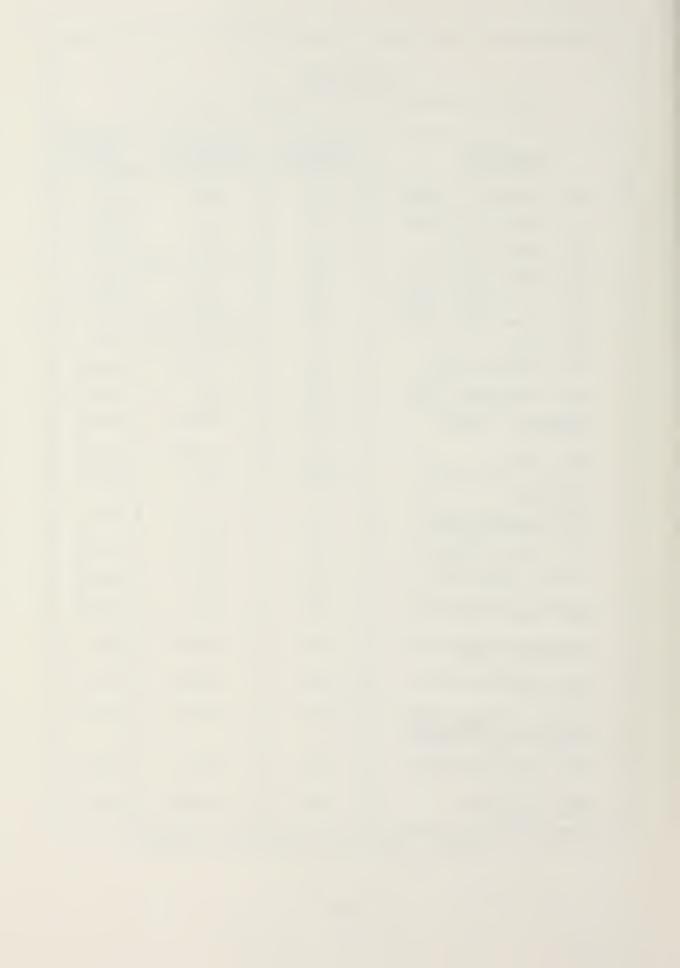


TABLE VII

Volume Minimization With Linear Variations

PARAMET ER	BASELINE RESULTS	OPTIMIZED RESULTS	CHANGE
	K 55 0 51 5	1130213	7,0
Total number of tubes	5230	5348	+2.3
% of tubes in cocler	7.0	7.3	+4.3
Tube length (ft)	10.3	8.7	-15.5
Tube inner diam. (in)	.527	*.729 .583	
Tube wall thick. (in)	.049	.022	- 55 . 1 (
Tube outer diam. (in)	.625	*.773 .627	
Tube pitch	1.4	*1.75 1.44	
Void diameter (ft)	1.10	1.28	+16.4
Bundle diameter (ft)	7.17	6.8	-5.2
Condenser volume (cu.ft.)	432.3	345.8	-20.0
Area ratio	0.266	0.298	+12.0
Coolant inlet vel. (ft./sec)	8.473	9.0	+6.2
Coolant mass flow rate (107 lbm/hr)	1.540	2.48	+61.0
Head loss (ft H2O)	7.35	5.84	-20.5
Pumping power (hp)	55.69	71.1	+27.7
Coolant temperature rise (°F)	9.90	7.1	-28.8
Log mean temperature difference (°F)	28.62	30.14	+5.3
Heat transfer area (sq.ft.)	8814.	7746.	-12.1
Average overall heat transfer coefficient btu/(hr-sq.ft0F)	574.9	713.1	+24.0
Steam exit-fraction (% of input)	13.3	0.5	-96.2
Heat rejected (10° btu/hr)	1.451	1.665	+14.7

^{*} Inner row values followed by outer row values.



VI. CONCLUSIONS

The intent of this research was to create a detailed condenser analysis code capable of being coupled with a numerical optimizer and to test the program to prove its versatility. An additional objective was to validate the analysis with existing data. The results of the test cases were presented in Chapter Five; the resulting confusions are summarized here.

were significant difficulties encountered formulating the complex condenser design analysis, CONDIP, in a way that was compatible with the optimizer COPES/CONMIN. However, the majority of those problems were overcome resulting in the creation of a program which, when combined with the optimizer, is capable of taking any initial design, no matter how impractical or infeasible, and solving for an solution based set of pre-determined on а constraints and design variables. There are still some minor limitations as to the degree of optimization, but the final design is usually within 10 pecent of the single best In addition, the test cases indicate that as many as ten design variables and six constraints can be simultaneously in the design optimization with CONDIP.

B. The test cases demonstrated the effectiveness of CONDIP as a design tool for not only the conceptual design of a condenser, but also in evaluating comparison studies based on any number of design variable combinations. The number of possible combinations of design objectives, design variables and design constraints implies limitless possibilities to be explored and evaluated.



- C. An attempt was made to verify CONDIP with existing data with inconclusive results. Part of the blame can be placed on the rather inadequate quality and quantity of the data, but the general performance of CONDIP's condenser indicates a weakness in the analysis. As stated earlier, the source of the this weakness may be found in the correlations used for condensate inundation. The constants used in the expression for correcting shell-side heat transfer coefficients are based somewhat on conjecture. Yet they play a significant role in the overall condenser performance. Despite this limitation, the ability to optimize CONDIP's detailed analysis is a significant step forward over using the traditional and limited HEI method.
- D. CONDIP incorporates features that further increase its appeal as a design tool. By possessing the ability to linearly vary pitch and tube diameter, a better understanding of how to improve condenser performance based on its configuration is realized. The capability to incorporate shell-side tube enhancement is another added plus. The possibilities that can now be investigated are limitless.



VII. RECOMMENDATIONS

In addition to the insight that this investigation has given into the generation of automated condenser design programs, it has specifically addressed the shortcomings and pitfalls which may be encountered along the way and offered possible solutions to overcome them. Presented below are recommendations for furthering the development of CONDIP as a completely versatile and accepted design program.

- A. Since the weak link and the most significant unknown in condenser analysis is the effect of condensate rain in typical condenser environments, subsequent research should be devoted to investigating this phenomenon and developing more precise analytic correlations. In particular, the effects of velocity and flow direction on the condensate film should be attended to.
- B. Perhaps in conjunction with the above, a test condenser should be constructed which is geometrically similar to the model proposed in CONDIP in order to physically observe and record the condenser performance. This data could then be used to either verify CONDIP or strengthen some of its analysis. In addition, this condenser should be built such that the tube bank can be arranged in any combination of pitch, tube diameter and row spacing to fully appreciate the effects of these variables.
- C. A series of sensitivity studies should be conducted on CONDIP to fully exercise its capabilities and determine the relative effects of various design variables on condenser performance. Tradeoff studies similar to those performed in this research would be most beneficial to fully understand condenser behavior.



- D. Additional subroutines should be created which would allow tube enhancement to be a design variable. This involves developing correlations between heat transfer enhancement and associated frictional losses. This type of relationship can be developed for both tube-side and shell-side enhancement.
- E. Finally, it is recommended that additional refinement be performed on the code to increase its capability and flexibility. One such way is to somehow allow pitch, tube diameter and tube wall thickness to vary linearly by row while still allowing the number of tubes to be a design variable. The options available are limitless.



APPENDIX A GLOSSARY

While it would most beneficial to present a complete glossary of all the variables used in CONDIP, the sheer number makes it difficult to present a comprehensive list. However, CONDIP makes liberal use of comment cards to define as many variables as possible to make the code easier to follow. Therefore, the computer listing in Appendix C is available for reference. A list of the possible design variables and constraints is provided here along with its corresponding position in the GLOBCM common block for easy reference in writing the appropriate COPES data cards. In addition, it will be specified whether these variables can be used as design constraints or design variables.

- 1. <u>ALST</u>: The length of the condenser and cooler tubes in feet. ALST is to be used only as a design variable.
- 2. <u>DELWP</u>: The pressure difference between the inlet and outlet coclant headers of the condenser bundle in psi. DELWP is to be used only as a design variable.
- 3. <u>DELWPC</u>: The pressure difference between the inlet and outlet coolant headers of the condenser bundle in psi. DELWPC is to be used only as a design constraint.
- 4. GFLOW: The mass flow rate of the coolant in lbm/hour. GFLOW cannot be used as a design variable simultanouesly with VELBI. Otherwise it can be used as a design variable or a design constraint.
- 5. SIDI: The tube inner diameter of the innermost row of



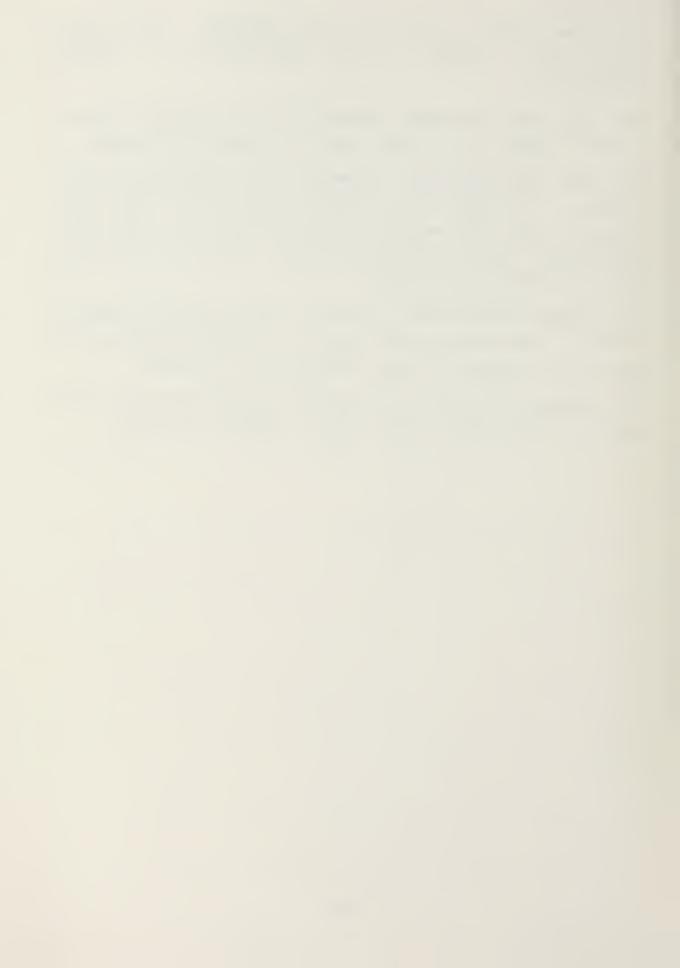
- the condenser bundle in inches. SIDI is to be used only as a design variable.
- 7. <u>SIDO</u>: The tube inner diameter of the outermost row of the condenser bundle in inches. If there is to linear variation of tube inner diameter then this variable represents the tube inner diameter of the entire condenser bundle. SIDO is to be used only as a design variable.
- 8. PHP: The coolant pumping power in horsepower. PHP is to be used only as a design constraint.
- 9. <u>RSPA</u>: The spacing between concentric rows in the condenser bundle in inches. RSPA is to be used only as a design variable.
- 10. <u>RADINS</u>: The inner void radius of the condenser bundle in feet. RADINS is to be used only as a design variable.
- 11. <u>REWI</u>: The tube-side Reynolds number of the coolant in the innermost row of the condenser bundle. REWI is to be used only as a design constraint.
- 12. REWO: The tube-side Reynolds number of the coolant in the outermost row of the condenser bundle. If there is no linear variation of tube inner diameter then this variable represents the tube-side Reynolds number of the entire condenser bundle. REWO is to be used only as a constraint.
- 13. <u>SDDI</u>: Tube pitch (tube spacing/tube outer diameter) of the innermost row of the condenser bundle. SDDI is to be used only as a a design variable.
- 14. <u>SDDO</u>: The tube pitch of the outermost row of the condenser bundle. If there is no linear variation of tube pitch then this variable represents the tube pitch for the entire condenser bundle. SDDO is to be used only as a design variable.



- 15. <u>SLDI</u>: Ratio of tube length to tube outer diameter of the outermost row of the condenser bundle. SLDI is to be used only as a design constraint.
- 16. <u>SLDO</u>: Ratio of tube length to tube outer diameter of the outermost row of the condenser bundle. If there is no linear variation of tube pitch then this variable represents the tube pitch for the entire condenser bundle. SLDO is to be used only as a design constraint.
- 17. <u>VELBI</u>: The velocity of the coolant in feet/sec. VELBI cannot be used as a design variable simultaneously with GFLOW. Otherwise it can be used as either a design constraint or as a design variable.
- 18. XW1: The ratio of tube thickness to tube inner diameter. XW1 can be used only as a design variable.
- 19. XW2: Tube thickness in inches. XW2 is to be used only as a design variable. XW2 and XW1 cannot be used simultaneously.
- 20. <u>VOL1</u>: The overall condenser and cooler volume in cubic feet. VOL1 is to be used only as a design constraint.
- 21. <u>VOL2</u>: The volume occupied by the tube bank, excluding the volume of the inner void, in cubic feet. VOL2 is to be used only as a design constraint.
- 22. <u>TNOTOT</u>: The total number of tubes in the condenser and cooler combined. If Option 1 is being used then TNOTOT is to be used only as a design constraint. If Option 2 is being used then TNOTOT is to be used only as a design variable.
- 23. <u>BNDRAD</u>: The condenser bundle in feet. BNDRAD is to be used only as a design constraint.
- 24. ARATIO: The ratio of the total cross-sectional area of



- the tubes (based on the tube outer diameter) to the tube sheet area. ARATIO is to be used only as a design constraint.
- 25. ODII: The tube cuter diameter of the innermost row in inches. ODII is to be used only as a design constraint.
- 26. ODOI: The tube cuter diameter of the outermost row in inches. If there is no linear variation of tube inner diameter then this variable represents the tube outer diameter of the entire condenser bundle. ODOI is to be used only as a design constraint.
- 27. <u>VLCMAX</u>: The maximum allowable steam velocity into the cooler. VLCMAX can be used only as a design constraint and only when a cooler is being designed in the system.
- 28. PRCCLR: The percent of the total number of tubes in the cooler. PRCCLR can be used only as a design variable.



APPENDIX B USERS MANUAL FOR CONDIP

This appendix describes the data cards that are necessary in order to couple any design program with COPES/CONMIN. Also described are cards illustrating data input required by CONDIP to initiate analysis. Thus, the data is divided into the COPES/CONMIN program section and the CONDIP-based condenser design program section.

The COPES data is segmented into "blocks" for convenience. All formats are alphanumeric for title, end and stop cards: F10 for real data; and I10 for integer data. The formatted input may be overridden by inserting commas between data entries. Comment cards may be inserted anywhere in the data stack prior to the end card and are identified by a dollar sign (\$) in column 1. The COPES data stack must terminate with an end card containing the word "END" in column 1-3. It should be noted that information pertaining only to single analysis and optimization is presented here. Information concerning the other options available in COPES along with further explanation of COPES capabilities can be found in [Ref. 13].

The analysis data is also segmented into blocks for convenience and they begin immediately following the "END" card in the COPES data. No comment cards are permitted here, and the analysis data stack must terminate with the word "STOP" in columns 1-4. This is where the initial design values are placed for entry into CONDIP.

Default values are recommended for use in the following COPES data cards unless otherwise noted. It is recommended that these values in the COPES data blocks be used until the user becomes familiar with the program. In addition a



sample data input is illustrated in figure B. 1 at the end of this appendix.



DATA BLOCK A

DESCRIPTION: COPES Title Card

FORMAT: 20A4

1	2	3	4	5	6	7	8
TITLE							

REMARKS:

1) This line is available for a brief description.



DATA BLOCK B

DESCRIPTION: COPES Program Control Parameters

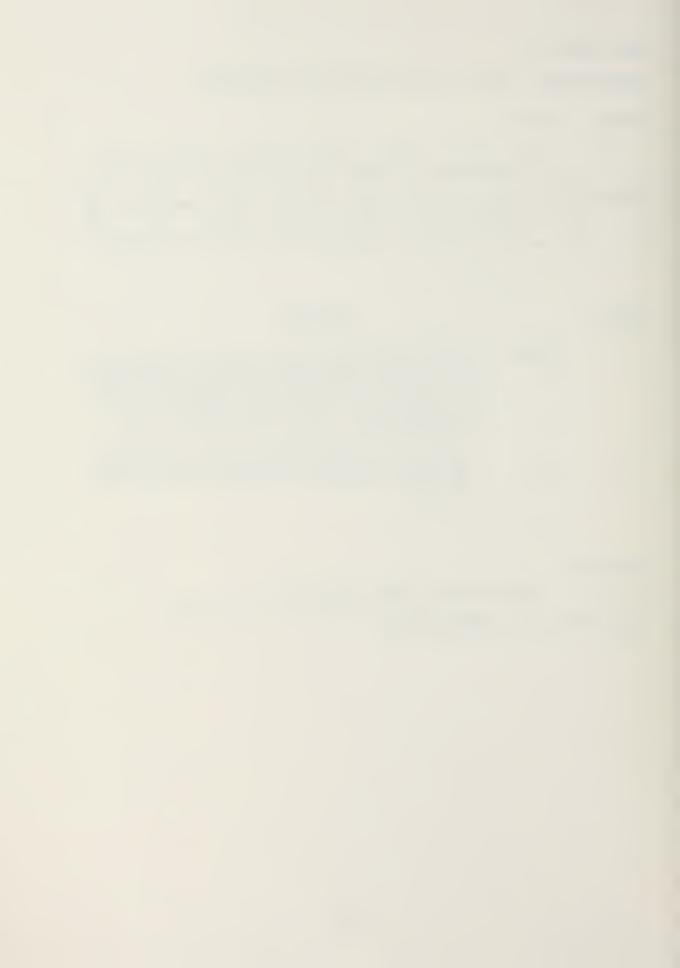
FORMAT: 7I10

1	2	3	4	5	6	7	8
NCALC	NDV						

FIELD		CONTENIS
1	NCALC: 0 1	Calculation control Read input and stop. Data of blocks A-B is required. Remaining data is optional. One cycle through the program. Data of blocks A-B is required. Remaining data is cptional. Optimization. Data of blocks A-I is required. Remaining data is optional.
2	NDV:	Number of independent design variables in optimization or optimum sensitivity study.

REMARKS:

- 1) Field 1 determines program execution
- 2) Fields 3-8 are to be left blank for the CONDIP application of COPES/CONMIN.



DATA BLOCK C

DESCRIPTION: COPES Integer Optimization Control Parameters

FORMAT: 8110

1	2	3	4	5	6	7	8
IPRINT	ITMAX	ICNDIR	NSCAL	ITRM	LINOBJ	NACMX1	NFDG

FIELD		CONTENIS
1	IPRINT: 0 1 2 3	Print control used in optimization program, CONMIN. No print during optimization. Print initial and final optimization information. Print above plus function value and design variable values at each iteration. Print above plus constraint values, direction vector and move parameter at each iteration. Print above plus gradient information. Print above plus gradient information. Print above plus ach proposed design vector, objective function and constraints during the one-dimensional search. required. Remaining data is optional.
2	IT MAX:	Maximum number of optimization iterations allowed. DEFAULT = 20.
3	ICNDIR:	Conjugate direction restart parameter. DEFAULT = NDV+1.
4	NSCAL:	Scaling parameter. GT.0 - Scale design variables to order of magnitude one every NSCAL iterations. LT.0 - Scale design variables according to scaling values input. DEFAULT = No scaling.
5	ITRM:	Number of subsequent iterations which must satisfy relative or absolute convergence criterion before optimization process is terminated. DEFAULT = 3.
6	LINOBJ:	Linear objective function identifier. If the optimization objective is known to be a linear function of the design variables, set LINOBJ = 1. DEFAULT = Non-Linear.
7	NACMX1:	one plus the maximum number of active constraints anticipated. DEFAULT = NDV+2.



DATA BLOCK C (Continued)

FIELD CONTENTS

8	NFDG:	Finite difference gradient identifier. All gradient information is computed by
	•	firite difference
	1	Gradient of objective is computed
		are computed by finite difference.
	2	Gradient of objective is computed analytically. Gradients of constraints are computed by finite difference. All gradient information is computed analytically

REMARKS:

- 1) The value of NSCAL = 0 is suggested and ITRM = NACMX1 = 0 should be used.
- 2) The value of IPRINT may be reduced when the user becomes familiar with the optimization output.
- 3) The default values will be used if the card is either left blank or a value of zero is entered.
- 4) Because of the complexity of the problem it is necessary to have a large value for ITMAX so the problem will not be terminated prematurely. Recommended value is ITMAX = 40
- 5) The complexity of the condenser analysis ensure that no function can be considered linearly dependent on any combination of variables. This justifies using the DEFAULT value for LINOBJ.



DATA BLOCK D

DESCRIPTION: COPES Floating Point Optimization Program
Parameters

FORMAT: 8F 10

1	2	3	4	5	6	7	8
FDCH	FDCHM	CT	CTMIN	C TL	CTLMIN	THETA	PHI

FIELD		CONTENTS
1	FDCH:	Relative change in design variables in calculating finite difference gradients. DEFAULT = 0.01
2	FDCHM:	Minimum absolute step in finite difference gradient calculations. DEFAULT = 0.001.
3	CT:	Constraint thickness parameter. DEFAULT = -0.1.
4	CTMI:	Minimum absolute value of CT considered in the optimization process. DEFAULT = 0.004
5	CTL:	Constraint thickness parameter for linear and side constraints.
6	CT LMIN:	Minimum absolute value of CTL considered in the optimization process. DEFAULT = 0.001
7	THETA:	Mean value of push-off factor in the method of feasible directions. DEFAULT = 1.0
8	PHI:	Participation coefficient, used if one or more constraints are violated. DEFAULT = 5.0.



DATA BLOCK D (continued)

FORMAT: 2F10

1	2	3	4	5	6	7	8
DELFUN	DABFUN						

CONTENTS

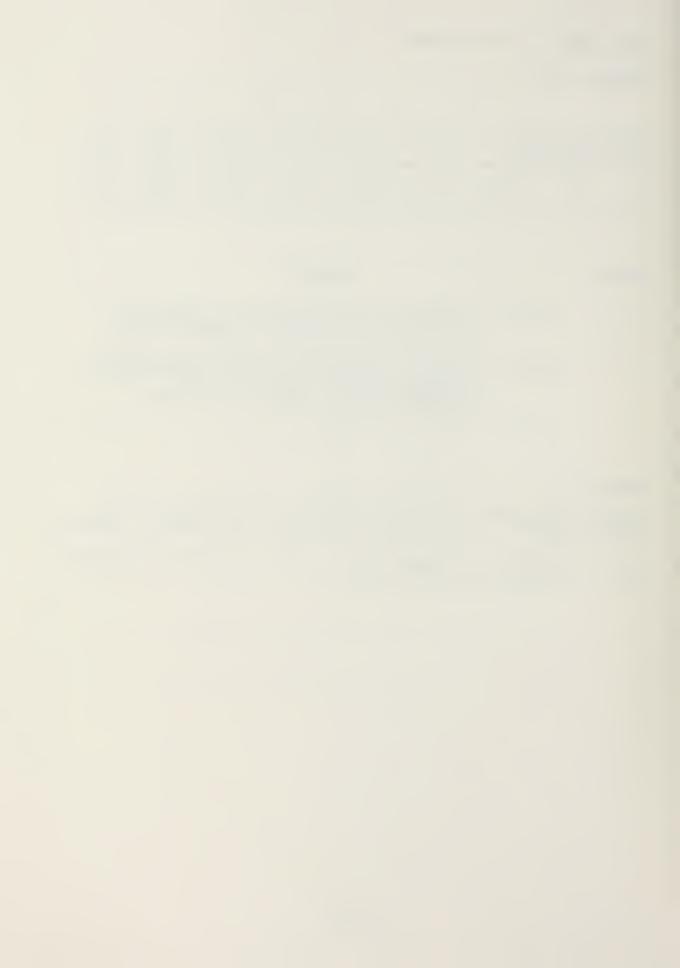
1 DELFUN: Minimum relative change in objective function to indicate convergence of optimization process. DEFAULT = 0.001.

2 DABFUN: Minimum absolute change in objective function to indicate convergence of the optimization process.
DEFAULT = 0.001 times the initial objective value.

REMARKS:

1) Note that data for Data Block D is entered on two separate cards. A blank card indicates the default value is to be used.

2) If the NDV is greater than 3, the recommended value for FDCH is between 0.05 and 0.10.



DATA BLOCK E

Total Number of Design Variables, Design Objective Identification and Sign on Design Objective. DESCRIPTION:

FORMAT: 2110, F10

1	2	3	4	5	6	7	8
NDVTOT	IOBJ	SGNOPT					

FIELD		CONTENTS
1	NDVTOT:	Total number of variables linked to the design variables. NDVTOT must be greater or equal to NDV. This option allows two or more parameters to be assigned to a single design variable. The value of each parameter is the value of the design variable times a multiplier which may be different for each parameter. DEFAULT = NDV.
2	IOBJ:	Global variable number associated with objective function in optimization or optimum sensitivity analysis.
3	SG NOPT:	Sign used on objective of optimization to identify whether function is to be maximized or minimized. +1.0 indicates maximization; -1.0 indicates minimization. DEFAULT = -1.0

REMARKS:

1) Currently there are not any variables in CONDIP which are linked to any of the design variables. Therefore the DEFAULT value is used for NDVTOT.



DATA BLOCK F

DESCRIPTION: Design Variable Bounds, Initial Values, and Scaling Factors.

FORMAT: 4F 10

1	2	3	4	5	6	7	8
VLB	VUB	X	SCAL				

FIELD		CONTENIS
1	VLE:	Lower bound on the design variable.
2	VUB:	Upper bound on the design variable.
3	X:	Initial value of the design variable. If X is non-zero, this will supercede the value initialized by subroutine ANALIZ.
4	SCAL:	Design variable scale factor. Not used if NSCAL 2 0 in Block C

- 1) There must be one separate data card for each design variable. Therefore there will be NDV data cards.
- 2) For all applications with CONDIP, initial values for the design variables will be entered through the INPUT subroutine called in ANALIZ.



DATA BLOCK G

DESCRIPTION: Design Variable Identification

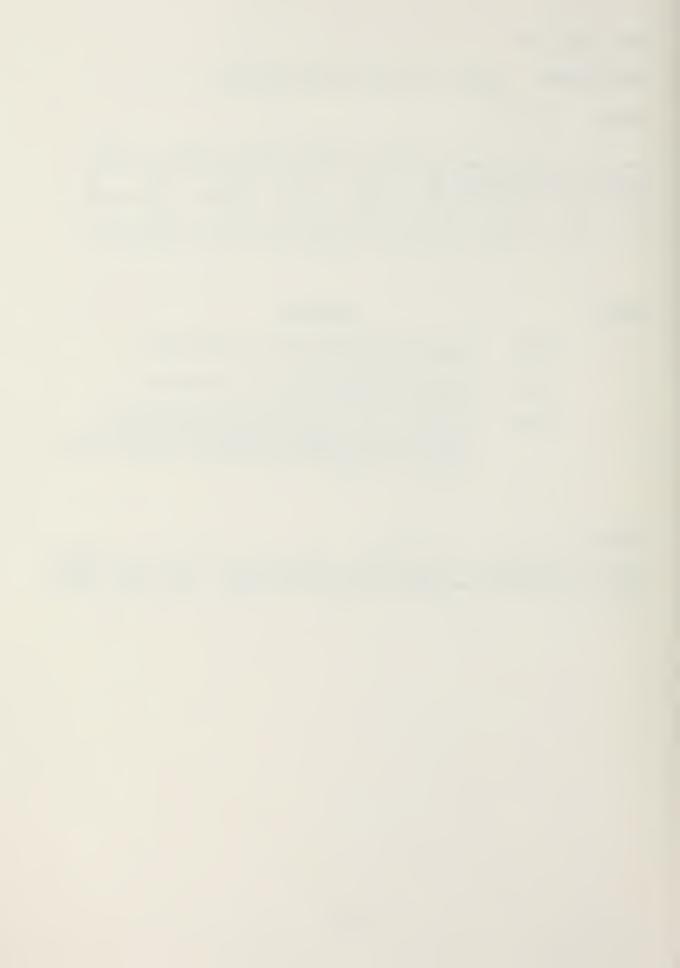
FORMAT: 2110, F10

1	2	3	4	5	6	7	8
NDSGN	IDSGN	AMULT					

FIELD		CONTENIŞ
1	NDSGN:	Design variable number associated with the variable.
2	IDSGN:	Global variable number associated with the variable.
3	AMULT:	Constant multiplier on the variable. The value of the variable will be the value of the design variable, NDSGN, times AMULT. DEFAULT = 1.0.

REMARKS:

1) There must be one separate card for each of the NDVTOT design variables. These data cards must follow the same order as the corresponding design variable parameter cards in Block F.



DATA BLOCK H

DESCRIPTION: Number of Constrained Parameters.

FORMAT: I10

1	2	3	4	5	6	7	8
NCONS							

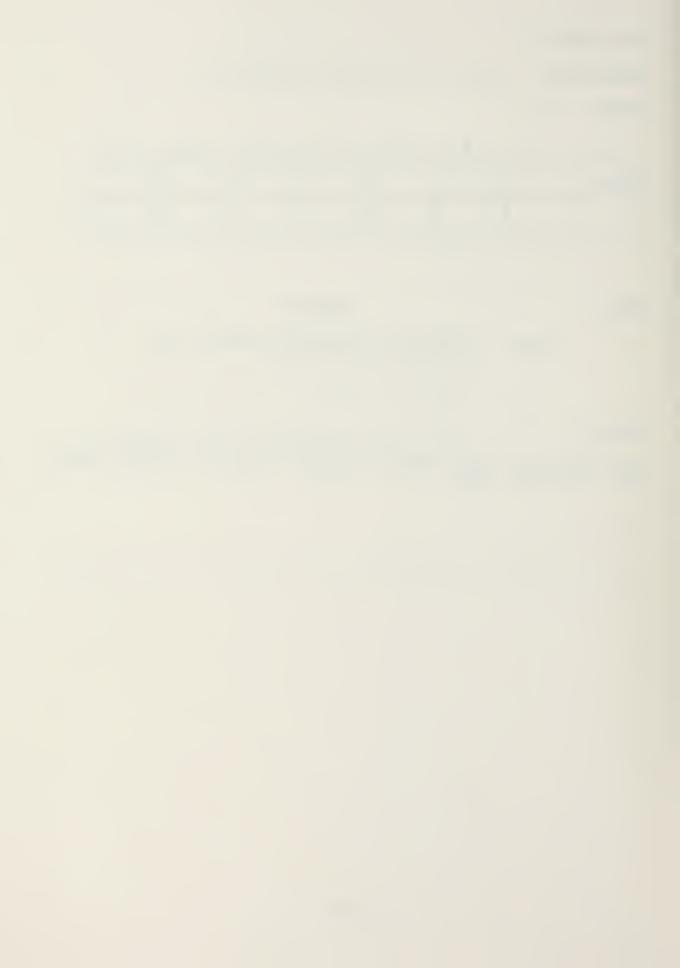
FIELD

CONTENIS

1 NCONS: Number of constraint SETS in the optimization problem.

REMARKS:

1) If two or more adjacent parameters in the Global common block have the same limits imposed, these are part of the same constraint set.



DATA BLOCK I

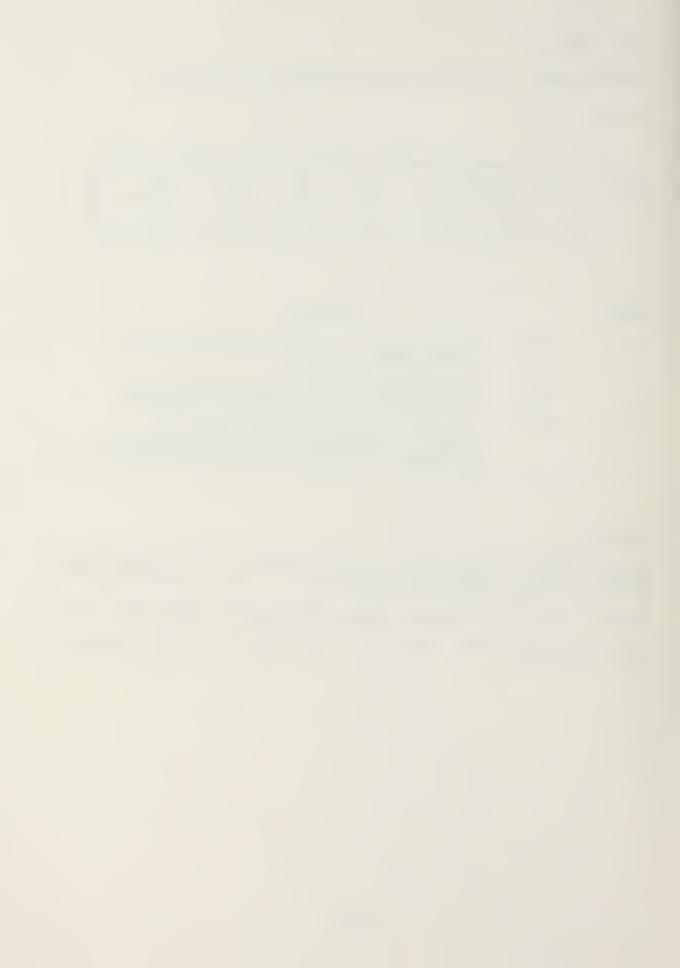
DESCRIPTION: Constraint Identification and Bounds.

FORMAT: 3110

1	2	3	4	5	6	7	8
ICON	JCON	LCON					

FIELD		CONTENTS
1	ICON:	First Global number corresponding to the constraint set.
2	JCON:	Last Global number corresponding to the constraint set. DEFAULT = ICON.
3	LCON:	Linear constraint identifier for this set of constrained variables. LCCN = 1 indicates linear constraints. DEFAULT = 0 = Nonlinear constraint.

- 1) In CONDIP there is only Global number and thus one constraint that comprise a constraint set. Therefore the DEFAULT value is used for JCON.
- 2) All the constraints in this analysis are nonlinear. The DEFAULT value was therefore used for LCON as well.
- 3) This is the first card of a two card set which must be read together.



DATA BLOCK I (Continued)

FORMAT: 4F 10

	1	2	3	4	5	6	7	8
В	L	SCAL1	ви	SCAL2				

<u>FIELD</u>		CONTENIS
1	BL:	Lower bound on the constrained variables. Value less than -2.0E+15 is assumed unbounded
2	SCAL1:	Normalization factor on lower bound. DEFAULT = Max of ABS(BL) or 0.1.
3	BU:	Upper bound on the constrained variables. Value greater than +2.0E+15 is assumed unbounded.
4	SCAL2	Normalization factor on upper bound. DEFAULT = Max of ABS(BU) or 0.1.

¹⁾ The normalization factor can usually be defaulted, with the notable exception of exit-fraction. the normalization factor used for this constraint is usually ten times the upper bound.



DATA BLOCK P

DESCRIPTION: COPES Data 'END' Card.

FORMAT: 3A1

1	2	3	4	5	6	7	8
EN D							

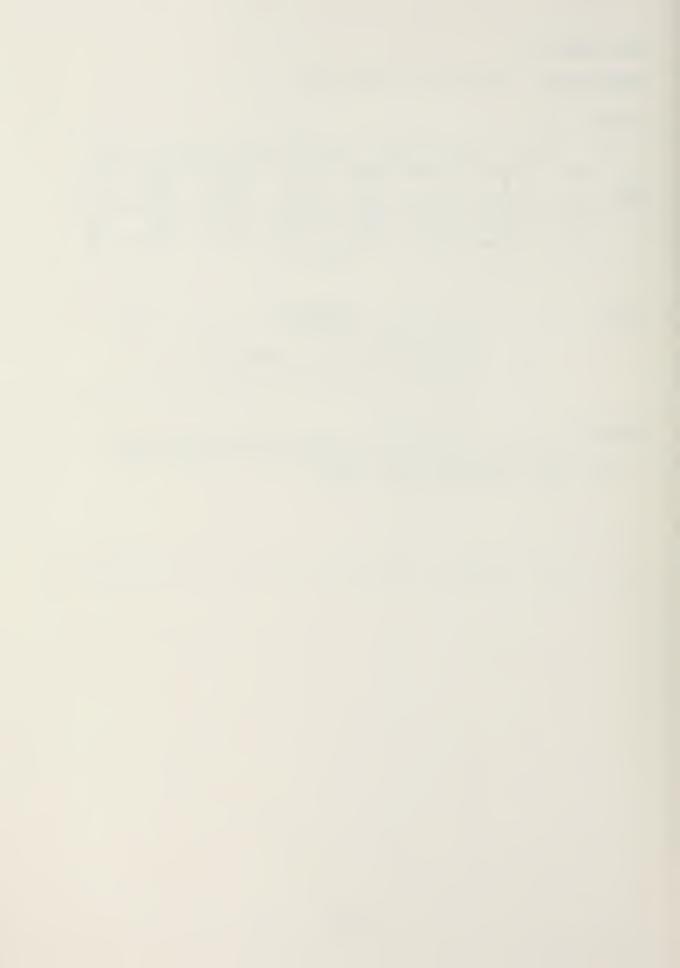
FIELD

CONTENTS

1

The word 'END' in column 1-3.

- 1) This card must appear at the end of the COPES data.
- 2) This ends the COPES input deck.



DATA BLOCK AA

DESCRIPTION: Geometry Option

FORMAT: 15

1	2	3	4	5	6	7	8
IOPT							

FIELD

CONTENTS

1	IOPT 1	Two condenser geometric options IOPT = 1. Number of condenser rows is a input as a constant; the number of tubes is a dependent variable. Use data blocks EF.FF.and GG.
	2	Use data blocks EE, FF, and GG. IOPT = 2. Number of tubes is allowed to be an independent variable and the number of rows is a dependent variable. Use data blocks HH and II. DEFAULT value is IOPT = 2.

REMARKS:

1) Data is right-justified and blanks will be interpreted as zeros.

2) If IOPT = 1, a smaller finite difference (FDCH) can be utilized in data Block D. This is because with this option the design analysis is less sensitive to the problems discussed earlier. Recommended using the DEFAULT value of 0.01 for FDCH.



DATA BLOCK BB

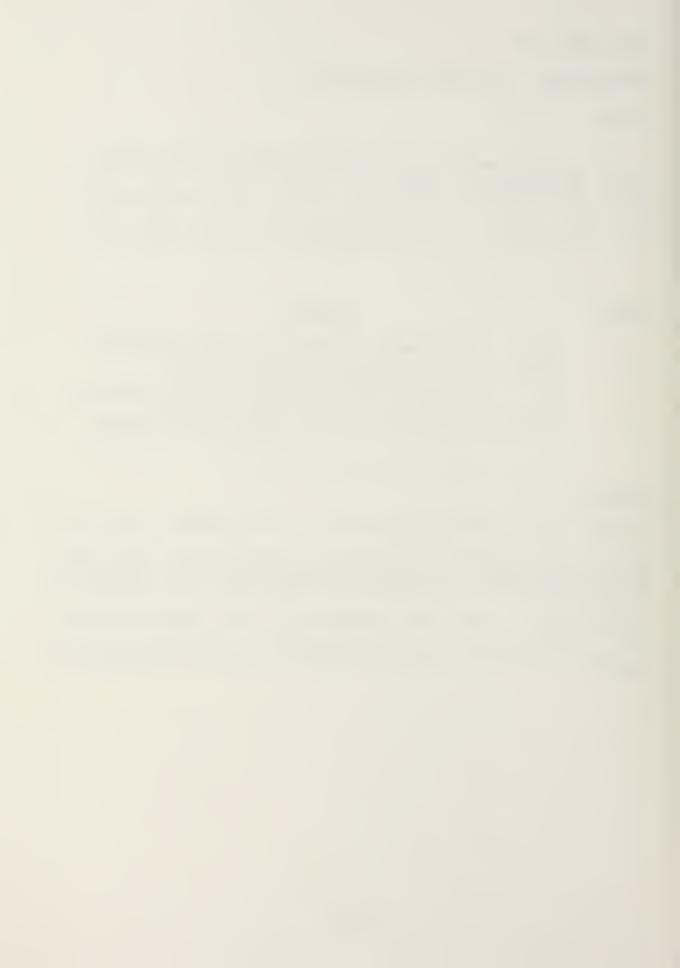
DESCRIPTION: Condenser Orientation

FORMAT: 115,3F10

1	2	3	4	5	6	7	8
ISEC	SECWID	PHI	PRCCL R				

FIELD		CONTENIS
1	ISEC	The number of sectors in the condenser.
2	SECWID	Sector width in degrees of arc.
3	PHI	Symmetry angle measure from the vertical.
<u>u</u>	PRCCLR	The percent of the tubes in the cooler.

- 1) Data BB is required, no matter what geometry option is chosen.
- 2) The only limitation on ISEC and SECWID is that their product is less than 360 degrees. If the product is exactly 360 degrees, certain trigonometric functions will return a singularity.
- 3) PHI is that angle from the vertical that cuts the condenser in half.
- 4) Data is right-justified and blanks will be interpreted as zeros.



DATA BLOCK CC

DESCRIPTION: Void Size, Tube Length, Row Spacing

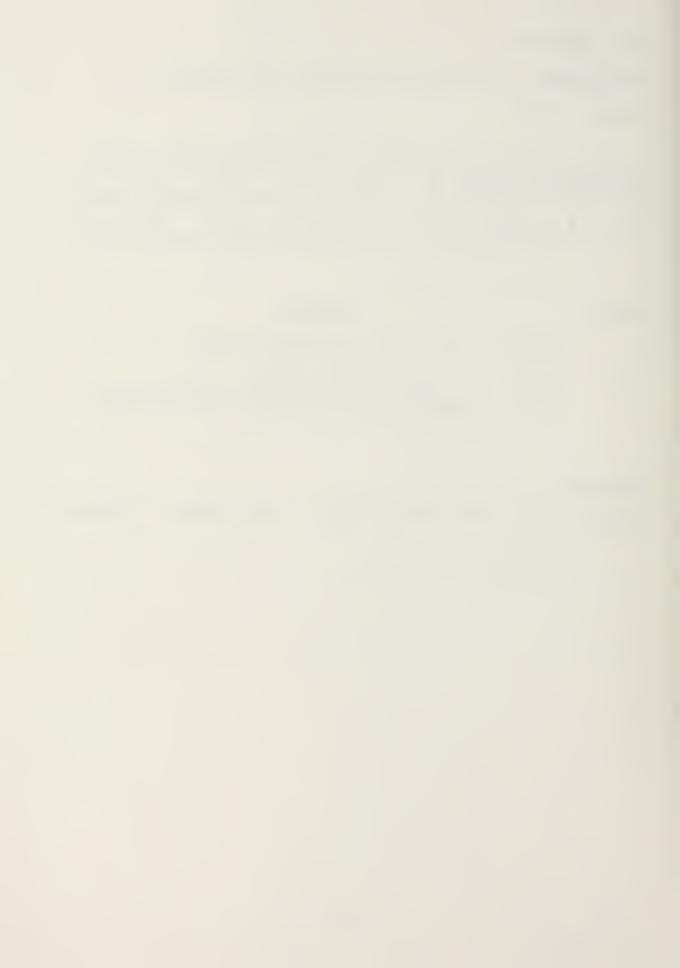
FORMAT: 3F 10

	1	2	3	4	5	6	7	8
R	ADINS	ALST	RSPA					

FIELD		CONTENIS
1	RADINS	The inner void radius; feet.
2	ALST	The tube length; feet.
3	RSPA	Concentric row spacing about the void; inches.

REMARKS:

1) Data CC is required, no matter what geometry option is chosen.



DATA BLOCK DD

DESCRIPTION: Tube Material Parameters

FORMAT: 15,4F10

1	2	3	4	5	6	7	8
IWALL	XW	TUBESW	SKW	FOUL			

FIELD		CONTENIS
1	IWALL 1 2	A Flag indicating the tube thickness specification. IWALL = 1. Tube thickness is input as ratio of tube thickness to tube inner diameter. IWALL = 2. Tube thickness is input in inches.
2	XW	The input for wall thickness, dependent on the value for IWALL.
3	TUBESW	Specifc weight of the tube material; lbm/(cu.ft.)
4	SKW	Tube material thermal conductivity; (btu-ft)/(sq.fthr-0F)
5	FOUL	Tube fouling factor.

- 1) Data DD is required, no matter what geometry option is chosen.
- 2) Data is right-justified and blanks will be interpreted as zeros.



DATA BLOCK EE

DESCRIPTION: Number of Rows

FORMAT: 15

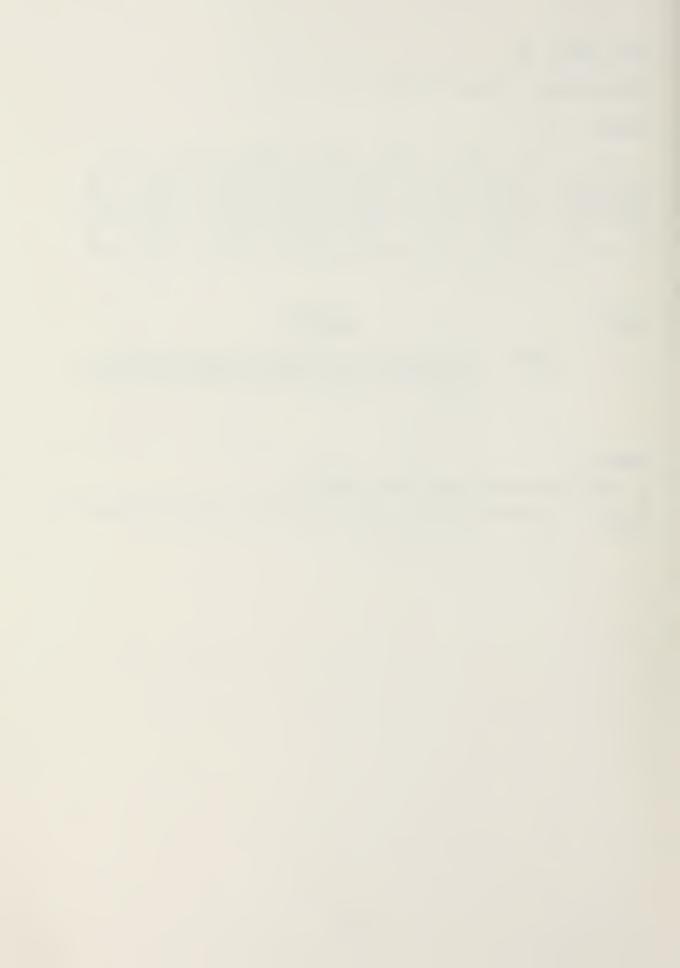
1	2	3	4	5	6	7	8
NOROWS							

FIELD

CONTENTS

NOROWS The number of concentric rows in the condenser bundle built around the center void

- 1) Data EE is used only when IOPT = 1
- 2) Data is right-justified and blanks will be interpreted as zeros.



DATA BLOCK FF

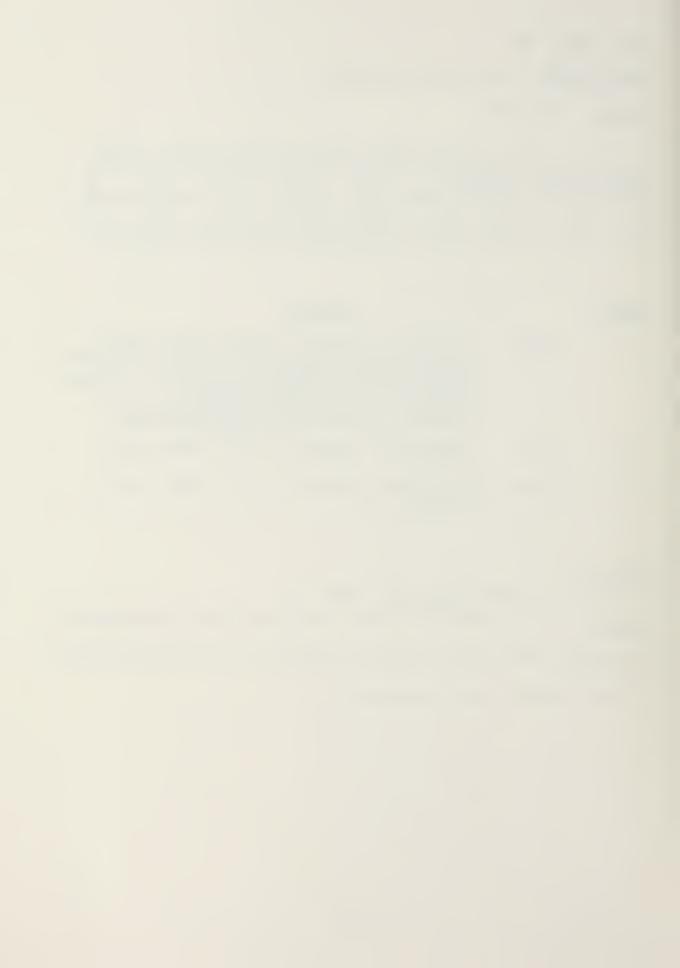
DESCRIPTION: Tube Inner Diameter

FORMAT: 115,2F10

1	2	3	4	5	6	7	8
MDIAM	SIDO	SIDI					

<u>FIELD</u>		CONTENIS
1	MDIAM	A flag to indicate whether tube inner diameter will linearly vary by row through the condenser bundle.
	2	MDIAM = 1. Tube inner diameter is uniform through the condenser bundle. MDIAM = 2. Tube inner diameter varies linearly through the bundle by row.
2	SIDO	Tube inner diameter of the outer row; inches.
3	SIDI	Tube inner diameter of the inner row; inches.

- 1) Data FF is used only when IOPT = 1
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) Ccoler tubes use the inner diameter of the innermost bundle row.
- 4) The DEFAULT value is MDIAM = 1



DATA BLOCK GG

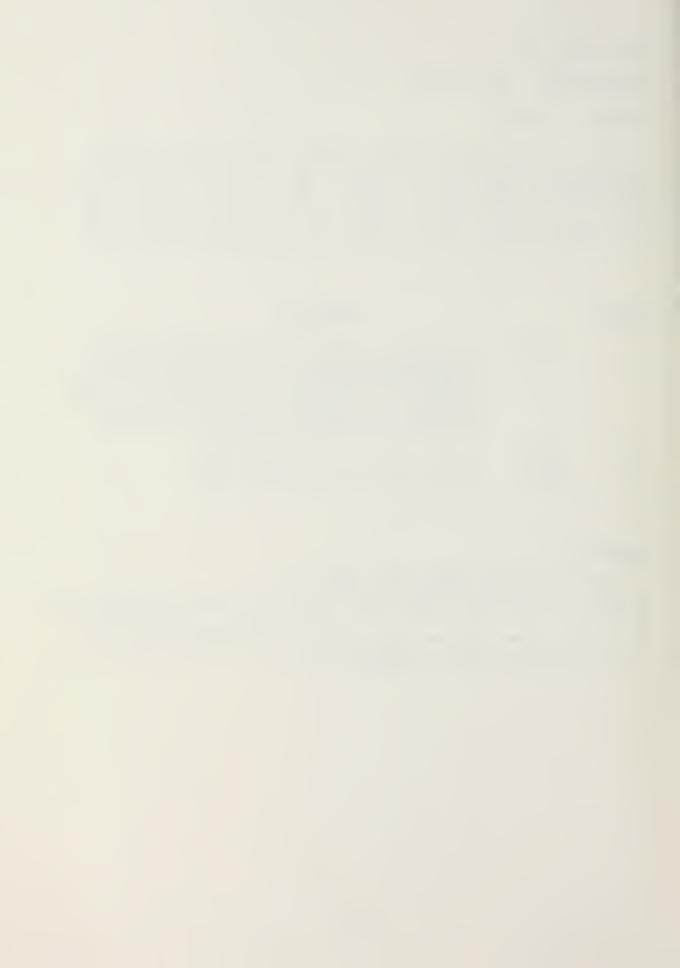
DESCRIPTION: Tube Pitch

FORMAT: 115,2F10

1	2	3	4	5	6	7	8
MPITCH	SDDO	SDDI					

FIELD	CONTENTS							
1	MPITCH	A flag to indicate whether tube pitch will linearly vary by row through the condenser bundle.						
	1	MPITCH = 1. Tube pitch is uniform through						
	2	the condenser bundla. MPITCH = 2. Tube pitch varies linearly through the bundla by row.						
2	SDDO	Tube pitch of the outer row;						
3	SDDI	Tube pitch of the inner row;						

- 1) Data GG is used only when IOPT = 1
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) Ccoler tubes use the pitch of the innermost bundle row.
- 4) The DEFAULT value is MPITCH = 1



DATA BLOCK HH

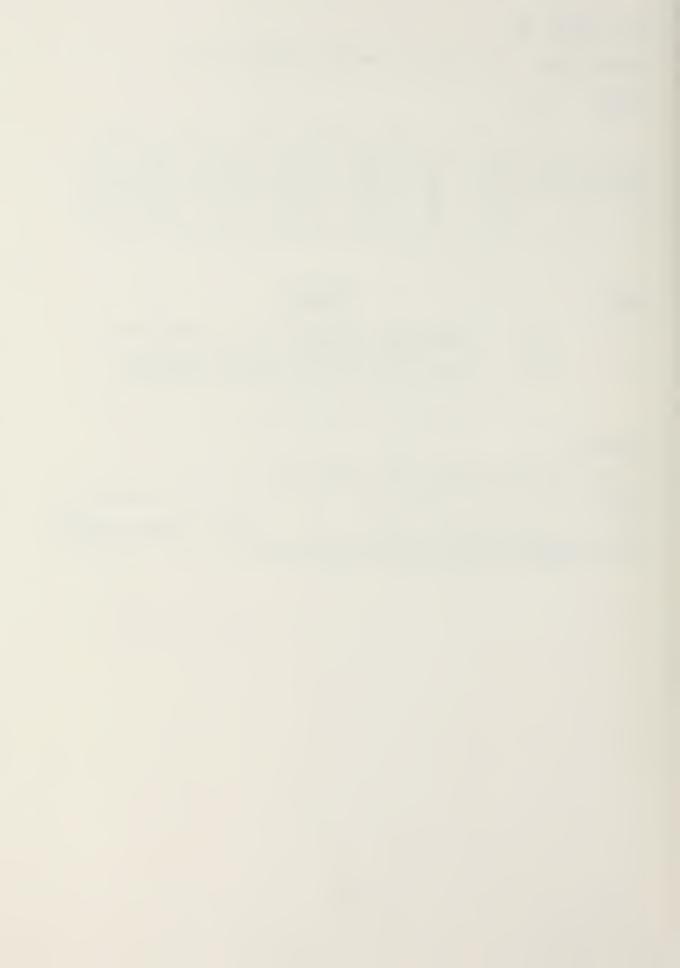
DESCRIPTION: Tube Inner Diameter and Tube Pitch

FORMAT: 2F 10

1	2	3	4	5	6	7	8
SIDO	SDDO						
	-						

FIELD		CONTENIS
1	SIDO	Tube inner diameter for the entire condenser; inches
2	SDDO	Tube pitch for the entire condenser

- 1) Data HH is used orly when IOPT = 2.
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) In the calculations, SIDI is set equal to SIDO and SDDI is set equal to SDDO. This avoids the need for two systems of nomenclature for each geometry option.



DATA BLOCK II

DESCRIPTION: Total Number of Tubes in the Condenser

FORMAT: F12

1	2	3	4	5	6	7	8
TNOTOT							

FIELD

CONTENTS

1 TNOTOT The total number of tubes in the condenser (cooler and the bundle).

- 1) Data II is used only when IOPT = 2.
- 2) Data is right-justified and blanks will be interpreted as zeros.



DATA BLOCK JJ

DESCRIPTION: Inlet Steam Mixture.

FORMAT: 15,2F10

	1	2	3	4	5	6	7	8
	JGAS	WSI	WNCIR					
Ì								

FIELD		CONTENIS
1	JGAS 1 2 3	A Flag in dicating the type of non-condensable gas entering the system. JGAS = 1. This indicates that the gas is air. JGAS = 2. This indicates that the gas is carbon dioxide. JGAS = 3. This indicates that the gas is a mixture of the two.
2 .	WSI	a mixture of the two. Steam flow rate entering the condenser; lbm/hr.
3	WNCIR	Ratio of the non-condensable gas flow to inlet steam flow; lbm/hr.

- 1) Data JJ is required, no matter what geometry option is chosen.
- 2) Data is right-justified and blanks will be interpreted as zeros.



DATA BLOCK KK

DESCRIPTION: Inlet Temperatures.

FORMAT: 2F10

1	2	3	4	5	6	7	8
STBI	STS AT1						

FIELD

CONTENTS

- 1 STBI Coclant inlet temperature; OF.
- 2 STSAF1 Inlet steam saturation temperature; or.

- 1) Data KK is required no matter, what geometry option is chosen.
- 2) Data is right-justified and blanks will be interpreted as zeros.



DATA BLOCK LL

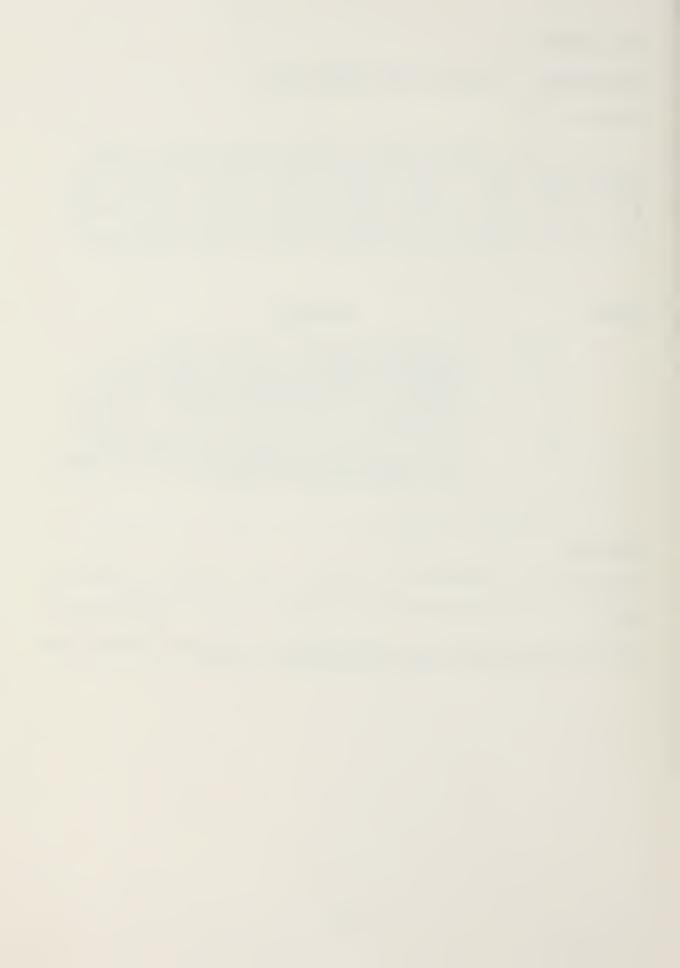
DESCRIPTION: Cooling Water Parameters

FORMAT: 2F 10

11	2	3	4	5	6	7	8
IFLOW	Х5						

FIELD		CONTENIS
1	IFLOW 1 2 3	A control flag for cooling water specifications. IFLOW = 1. Input pressure drop across cooling water headers in psia. IFLOW = 2. Input cooling water velocity in ft/sec. IFLOW = 3. Input coolant flow in lbm/hr.
2	x 5	Actually input the value for flcw into this variable. The specification for flow to be determined by IFLOW

- 1) Data LL is required, no matter what geometry option is chosen.
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) X5 acts as a temporary all-purpose storage variable for whatever expression for coolant flow is used.



DATA BLOCK MM

DESCRIPTION: Internal Enhancement Regions

FORMAT: 15

1	2	3	4	5	6	7	8
NEI							

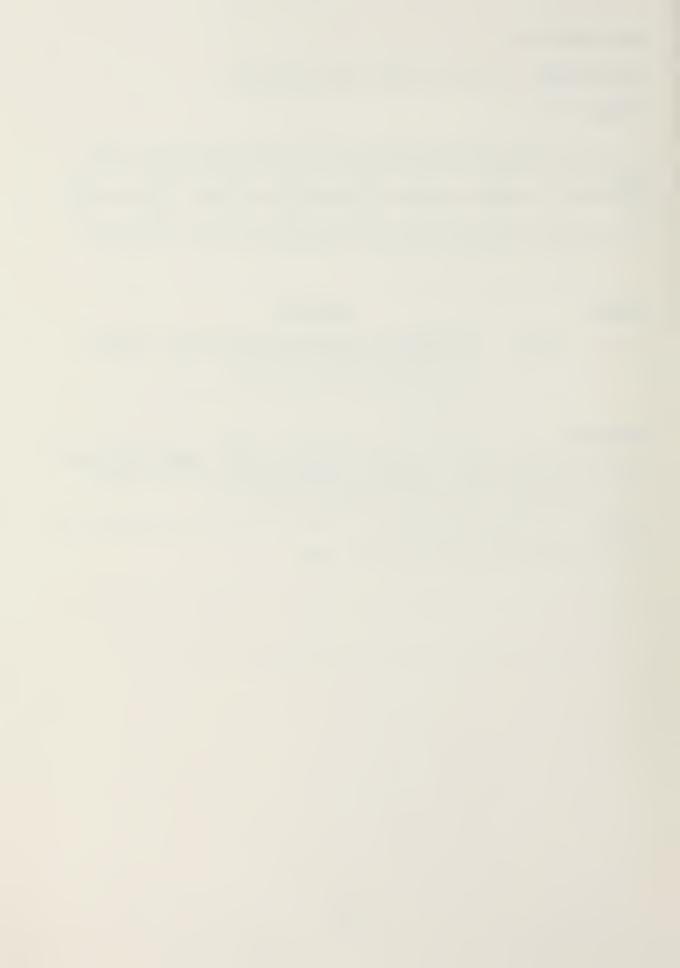
FIELD

CONTENTS

1 NEI

Number of internal enhancement regions. A value between 1 and 6.

- 1) Data MM is optional. There must be NEI subsequent data cards providing the necessary parameters for each region.
- 2) enhancement can only be used if IOPT = 1.
- 3) Data is right-justified and blanks will be interpreted as zeros.
- 4) This value was zero for all runs.



DATA BLOCK NN

DESCRIPTION: Internal Enhancement Parameters

FORMAT: 215,3F10

11	2	3	4	5	6	7	8
NRNI	NETI	ENI	BC	BF			

FIELD		CONTENIS
1	NRNI	Row number of first row in internal enhancement region.
2	NETI	Number of tubes in each internal enhancement region.
3	ENI	Internal heat transfer enhancement factor.
4	BC	Coefficient in internally enhanced tube coclant pressure drop calculation.
5	BE	Exponent in coolant pressure drop calculation.

- 1) Data NN is optional. However, if NEI is greater than zero then there must NEI "NN" data cards to provide the necessary data for each enhancement region.
- 2) enhancement can only be used if IOPT = 1.
- 3) Data is right-justified and blanks will be interpreted as zeros.
- 4) These values are constant for entire run and cannot be changed by the optimizer.
- 5) This value was zero for all runs.



DATA BLOCK OO

DESCRIPTION: External Enhancement Regions

FORMAT: 15

1	2	3	4	5	6	7	8
NEE							

FIELD

CONTENTS

1 NEE

Number of external enhancement regions. A value between 1 and 6.

- 1) Data 00 is optional. There must be NEE subsequent data cards providing the necessary parameters for each region.
- 3) Data is right-justified and blanks will be interpreted as zeros.
- 4) This value was zero for all runs.



DATA BLOCK PP

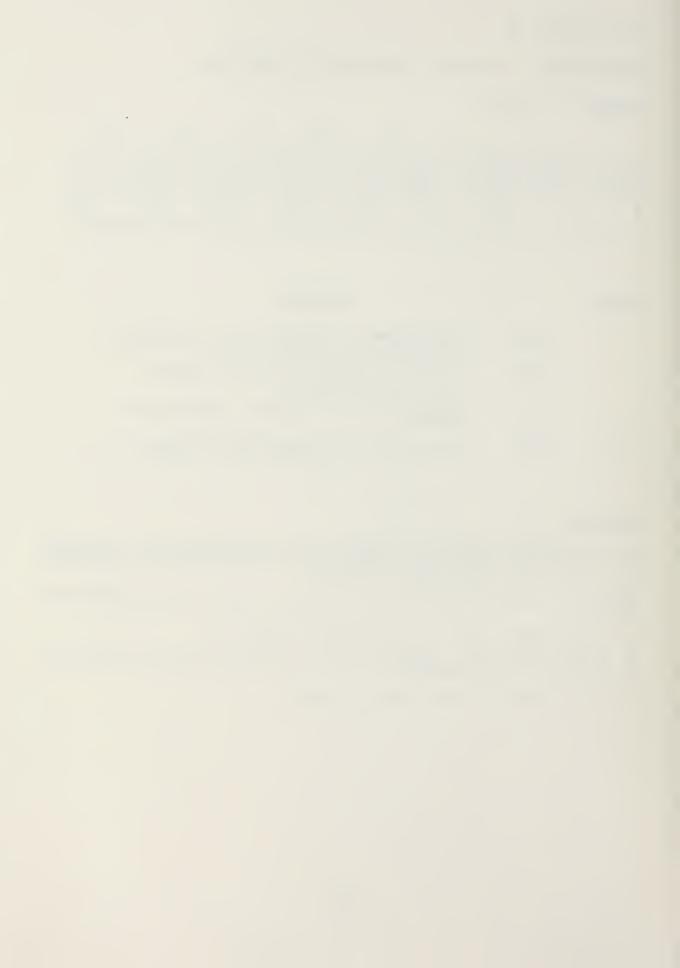
DESCRIPTION: External Enhancement Parameters

FORMAT: 215,2F10

1	2	3	4	5	6	7	8
NRNE	NETE	E NO	ENH				

FIELD		CONTENIS
1	NRNE	Row number of first row in external enhancement region.
2	NETE	Number of tubes in each external enhancement region.
3	ENI	External heat transfer enhancement factor.
4	ENH	Steam-side pressure drop factors.

- 1) Data PP is optional. However, iF NEE is greater than zero then there must NEE "PP" data cards to provide the necessary data for each enhancement region.
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) Enhancement can only be used if IOPT = 1.
- 4) These values are constant for an entire run and cannot be changed by the optimizer.
- 5) This value was zero for all runs.



DATA BLOCK QQ

DESCRIPTION: Baffle Options

FORMAT: 15

 1	2	3	4	5	6	7	8
IBAF							

FIELD

CONTENTS

1 IBAF

A flag to be used to determine baffle number and location.

- 1) Data QQ is optional.
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) These values are constant for an entire run and cannot be changed by the optimizer.
- 4) Additional specified baffles were not used in any of the runs. This value was zero for all runs.



DATA BLOCK RR

DESCRIPTION: Baffle Location

FORMAT: 15

	11	2	3	4	5	6	7	8
	JBAF							
I								

FIELD

CONTENTS

1 JBAF An array containing baffle locations

- 1) Data RR is optional.
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) These values are constant for an entire run and cannot be changed by the optimizer.
- 4) Additional specified baffles were not used in any of the runs. This value was zero for all runs.



DATA BLOCK SS

DESCRIPTION: Detailed Printout

FORMAT: 15

1	2	3	4	5	6	7	8
IPRT							

FIELD

CONTENTS

1 IPRT

A flag to generate a detailed output of the condenser analysis (OUT3)

- 1) Data SS is optional.
- 2) Data is right-justified and blanks will be interpreted as zeros.
- 3) These values are constant for an entire run and cannot be changed by the optimizer.
- 4) This value was zero for all runs.



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VOLUME MINIMIZATION - CIRCULAR CONDENSOR 2,8 1,40,,5,,,20 .10
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1000..8000.

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.022,.109...

.407,1.206...

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527 1.40 282.

161 970. 0000371

76.66 110.52
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```

Pigure B.1 Sample Data Input.



APPENDIX C CONDIP LISTING

The following Appendix contains a complete listing for CONDIP. An effort has been made to make the program as readable as possible through liberal use of comment cards.



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E MATERIAL LBM/CO
EC
TEAM MASS FLOW
FELOUR = TUBE FOULTING FACTOR.

JEAGH = CCONTROL FLAG FOR COOFLING WATER

IFLOR = CCONTROL FLAG FOR COOFLING WATER

IFLOR = CONTROL FLAG FOR COOFLING WATER

ISEC. = CONTROL FLAG FOR COOFLING WATER

JAS = NCONTROL FLAG FOR TUBE THICKNESS SPECIFICATI

JAS = NCONTROL FLAG FOR LINES THICKNESS SPECIFICATI

JAS = NCONTROL FLAG FOR LINES VARIATION IN TUBE FOR MAINTENANCE MENT

NEI = NUMBER OF EXTERNAL ENHANCEMENT REGIONS.

NEI = NUMBER OF TUBES IN THE CONTROL FRAME ENHANCE MENT

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FLCW-EQ-1) DELWP=X5
FLCW-EQ-2) VELBI=X5
FLCW-EQ-3) GFLCW=X5
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NEI-ET-0) AND (NEI-GT-6)) GO TO 120
1=1,NEI NEI CT (1), BE(I)
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1 BAF
- 2). OR. (IBAF. GT. ISEC)) GO TO
0). OR. (IBAF. EQ.-1)) GO TO 80
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AD (5,400) IWALL, XW, TUBESW, SKW, FOUL
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PITCH, SDDO, SDDI
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(1H0,38H****
(1H ,43 HRAN GE OF ALLOWABLE ISEC IS 1 THRU 15, ISEC=,15)
(1H ,60HCON DENSER ANGULAR WIDTH IS GUT OF RANGE 0 TO 360
'H=,E10,5,5H DEG.)
(1H ,5HISEC=,15,10H SECWID=,E10,3)
(1H ,5HNUMBER OF ROWS MUST BE IN RANGE 0 TO 100, NOROWS:
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NRNE(6), NRNI(6), NR
                     A ALST, DELWP, DELWPC, EWO, SDDI, SDDO, SLDI, SLDI, SDDC, VLCI BC(6), BE(6), ENH(6), EN NCIR, IOPT, PT ST (15), SE NEI, NET F (6), ST
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10.2, 14x, 6HINCHES,
5 MPITCH =, 12,
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INSIDE RADIUS, 12X, F10, 4, 16X

TUBE OD VARIATION IS MOIA

IDE DIAMETER, 9X, F10, 4, 14X, 10E

IDE DIAMETER, 9X, F10, 4, 14X, 10E

IDE DIAMETER, 13X, 11HOU

IDE DIAMETER, 15X, 13X, 11HOU

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              11X, F13.5/)
E,9X, F9.0,16X,6HLBM/HR
N/C FLOW,3X,1PE13.5/1
                                                                                                                                                                                                                           \overline{\mathsf{x}}
                                                                             15X,28HCW FLOW SPECIFIER IS IFLCW = 12)

22X,22HCOOLANT MASS FLOW RATE, 15X, F9.1, 15X, 6HLBN-15X, 16HCOOLANT VELOCITY, 14X, F9.1, 15X, 6HFT/SEC/)

15X,21HCOOLANT PRESSURE DROP, 9X, F11.3, 16X, 3HPSI, 15X, 38HNO. OF INTERNAL ENHANCEMENT REGIONS IS, IS, 15X, 26HFLAG FOR BAFFLE IS IBAF = , 15)
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            15 X,19 HTUBE FOUL ING FACTOR, 11 X, F13.;
15 X,20 HSTEAM MASS FLOW RATE,9X, F9.0;
15 X,27 HWEIGHT FRACTION OF N/C FLOW;
15 X,25 HN/C GAS IS CARBON DIOXIDE;
15 X,25 HN/C GAS IS A MISTURE OF AIR A
15 X,29 HCOOLANT INJECTION TEMPERATURE
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3.W SI.PFILL
COMMON /ORC1/ AOTFLW(100), SID(100), TBNPR(100), AULFLW(100), CN
10)
COMMON /ORC2/ ANT(100, 15), STB2ES, VW(100)
COMMON /OUT/ DELOD, SMTB1, SMTB2, SMWB, SUMQ, TNO, VEL2, FR
COMMON /CONST/ AMWNC, CB1, CPB, P1, SG, IFIRST
COMMON /SEC/ ALMTD(100, 15), CUMDP(100, 15), DELP(15), GFLW(100, 15), FR
100, 15), PMIX(100, 15), PSAT(100, 15), OBELP(15), GFLW(100, 15), FR
20, 15), SHI(100, 15), SHN(100, 15), WCND(100, 15), WGAS(15), WP(15), WSL(15), WP(15), WSL(15), WP(15), WSL(15), WP(15), WP(15), WSL(15), WP(15), WSL(15), WP(15), WSL(15), WSP(15), WSP(1
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IF+XW2/6.
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IF+(2*SIDIF*XW1)
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DDF+XW2
IDOF+(2*
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TNR=[ARCPR*RADIUS(I))/CSPU
TNOS=TNOS+TNR
IN OC=0.

TN OS=0.

TN OS=0.

TN R=0.

TD IFF=0.

SI DOF= SIDO /12.

SI DIF=SIDI /12.

IF (XWZ.01=0) DDDF=SIDDF+X

IF (XWZ.01=0) DDDF=SIDDF+X

IF (XWZ.01=0) DDDF=SIDDF+X

IF (XWZ.01=0) DDDF=SIDDF+X

OD I=0 COF*12.

DE LOD F=0.

DE LOD F=0.

DE LOD F=0.

DE LOD F=0.

SE PP= S DD + DDDF

CS PI=S DD + DDF

CS PI=S DD
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THE DIMENSIONS OF NUMEROUS
                                                                                                                                   GEOMETRY
                                                                                                                                   PERTINENT
                                                                                                                                                                                                                                                                                                                                                           AO TFLW (NOROWS+1)=ALST *(RADINS-SDI*ODII/24.) *PI/6.
                                                                                                                                              SD DL= (RADIUS(I)*AR CPR)/(OD OF*TBNPR(I))
CS PL= SDCL*ODOF
TP AR(I)=1./(CS PL*R SPF)
AD LFLW(I)=(CSP L-OD OF)*TBNPR(I)
AO TFLW(I)=AOLFLW(I)*ALST
TN OS= TN CS+TBNP R(I)
CONTINUE
AH OLE S=PI*(OD OF**2)*TN O/4.
RO WS=FLGAT(NOR OWS)
BN DRAD=RADINS+R SPF*(ROWS-I.)
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(SID,NOROWS)
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(AOLFLW,NOROWS)
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AO LFLW(I)=(CSPO-DDOF)*TNR
AO TFLW(I)=AOLFLW(I)*ALST
TBNPR(I)=TNR
RADIUS(I+I)=RADIUS(I)+RSPF
I= I+1
NO ROW S=NOROWS+1
                                                                            IF (ND ROWS LT 100) GD TO
ERRI=TNO-(TNOS*SECFLG)
TN D=T N DS*SECFL G
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NO ROWS = NO ROWS - 1
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AR ATI O=AHOLES/TSAREA
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                                                         )+SECWID*(I-1)
SECA=SECA-360.
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                                                                                                                                    DUMMY=ALST*PI*SECWID*SECFLG/360.
BNDRAI=BNDRAD
VGLI=DUMMY*BNDRAI**2
VOLIC=VGLI
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MUST BE A
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SHEET AREA
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                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               •1 GO TO 150
IF (I SEC.EQ.1) GD TO 110
FCWID=SECWID*(SECFLG-1.)
DO 110 I=1 ISEC
SECA=(PHI-FCWID/2.)+SECWI
IF (SECA.GT.360.) SECA=SE
SECANG(I)=SECA
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  THIS SECTION FOR
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RCORR=RSPF
RCORR=RCORR*(I.-PFILL))
BN DRAD=BNDRAD-RCORR
VOL1=DUMMY*BND RAD**2
VOL1C=VCL1
                                                                                                                                                                                                                                OUTSIDE BUNDLE RADIUS
INCREASE OR DECREASE
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(6,700) AHOLES, TSAREA, ARATIO
(6,710)
[=1,000 DWS
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(6,720)
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IF (IBAF) 160, 200, 180

IBAFT = ISEC+1

DO 170 I=1, IBAFT

JBAF(I) = I
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AN = SUMMING RAGISTER FOR NO. OF TUBES

AN T = NO. OF TUBES OVER THE TARGET TUBE

BAF = ANGLE ABOUT HORIZONTAL OF 1ST BAFFLE ABOVE CURRENT SECTORCONO

DELOF = SEE ABOUT HORIZONTAL OF 1ST BAFFLE ABOVE CURRENT SECTION

ON CONO

IS = SHORTEST DIST FROM HORIZONTAL TO CURRENT ROW INTERSECTION

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IS = SHORTEST DIST FROM HORIZONTAL TO CURRENT ROW INTERSECTION

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AL = COEF IN FF CALCULATION

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CB I = COOLANT SALT CONCENTRATION IN WT PROM

CM DOT = COOLANT MASS FLOW THROUGH CONDENS ER LBM/SEC

CM TOT = TOTAL COOLANT MASS FLOW THROUGH CONDENS ER LBM/SEC

CM TOT = TOTAL COOLANT MASS FLOW THROUGH CONDENS ER LBM/SEC

CM TOT = TOTAL COOLANT MASS FLOW THROUGH CONDENS ER LBM/SEC

FF = TUBE FRICTION FACTORS

HEAD = FT**2/SEC***

SID = STORED INNER TUBE DIAMETER

SID = STORED INNER TUBE INNER DIAMETER! RATIO

SLD = STORED (LENGTH/TUBE INNER DIAMETER! RATIO

SLD = SLD RATIO OF TUBES IN OUTER ROW
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AN = AN+ (2T-D1) / (RSP F*2.)

IF (DRCA7.EQ.1.) WRITE (6

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II = ITR-12+1

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D2 = SQRT (R1*R1-XCS)

IF (DRCA7.EQ.1.) WRITE (6

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PART - 1
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VEL C(100), PMIX
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FFC(100), UNC (10
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 4), WSP (15)
COMMON /CONST/
COMMON /COOLI/
I), ALM TOC(100), S
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COMMON /COOL/
1 BZC, ENHIC, ENHOC
DI MENSICO (15)
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OCCOUNTIES DE LA TRANSPORTICA DE L



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NUMBER OF VERTICAL AND HERIZGNTAL ROWS IN THE COOLER
ONE
                                                                                          /SEC
           FT ** 3/L BM: T ** 3/L B M
R
STSATI = ENTERING MIXTURE SPECIFIC VOL FT**3/LBM SVNC1 = ENTERING MIXTURE SPECIFIC VOL FT**3/LBM SVNC1 = ENTERING N/C GAS SPECIFIC VOL FT**3/LBM TSATI = ENTERING STEAM SATURATION TEMP R WGAS = N/C FLOW TO ENDENSER LBM/HR WNCI = N/C GAS FLOW OF N/C TO MASS FLOW OF N/C SECTOR LBM/HR WS I = TOTAL STM FLOW TO CONDENSER LBM/HR VELO I = MIXTURE VELOCITY IN FIRST ROW OF TOBES FILLING FIRST ROW OF TOBES FILLING FIRST ROW OF TOBES IN THE COOLE IHNOC = THE NUMBER OF VERTICAL TOBES IN THE COOLE HTCLR = THE HEIGHT OF THE COOLER FT**3/LBM
                                                                                                                                                                                                                                                                                                                                                                                                                                      0
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                                                                                                                                                                                                                                                              SD DC= SDDI
TN DC= (PRCCLR/(100.-PRCCLR))*TND
HTCLR=BNDRAD-RADINS
VN DC= (HTCLR/(SDDC*DDIF*O.886))+1.
IHNDC= TNOC/VNDC
HN DC= FLOAT(IHN DC)
VN DC= TNOC/HNDC
IVNOC= VNDC+.5
VN DC= FLOAT(IVN DC)
                                                                                                                                                                                ROWS=FLOAT(NORGWS)
TN OC=0.
IV NOC=1
IF (PRCCLR.EQ.0.0) GO TO 10
                                                                                                                                                                                                                                                                                                                                                                 TOP = NCROWS+1
CFLG=FLOAT(ISEC)
CI=WSI*WNCIR
LOOP=0
AVE=(WSI+WNCI)/SECFLG
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IF THERE
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SECTOR
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AMOLST*VGI I+AMOLNC/ (SVNCI*AMUL ST)
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AR AT = A OTFLW(1) / SECFAR
VE L1 = (WS1+WNCI) *SVMIXI/(3600.0*SECFAR*SECFLG)
VE L2 = VE L1/ARAT
VE L2 = VE L1/ARAT
IF (AR AT -0.715) 30.30.40
DE LP12 = 0.4*(1.25-ARAT)*VEL2**2/G2
GO TO 50
DE LP12 = 0.75*(1.0-ARAT)*VEL2**2/G2
DE LP12 = 0.75*(1.0-ARAT)*VEL2**2/G2
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ST BIR = STB I +459 . 69
PT LIM = PSATFN(STBIR)
TS ATI = STSATI +459 . 69
PS ATI = STSATI +459 . 69
PS ATI = PSATFN(T SATI)
AM OLN = WNC I / AM WNC
HF G = HF G F N (STSATI)
AM OLS S = WSI / 18 . 015
AM OLS T = AMOL SS + AMOL SY
AM OLS T = AMOL SS + AMOL SS / AMOL ST **V SW M S = 0 . 0
SM M B = 0 . 0
SM M B = 0 . 0
SM M B = 0 . 0
SW M B = 0
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SECTOR LOOP
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                                                      MOLNC, AMOLSS, AMOLST
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                                                                                                                                         ON SECTORS
MAIN
      S(1,1), WGAS(I)
                  HFG, AOTFLW(I
                                                                                                                  ELP 12, DELP2
                              SAT 1, STSAT 1
                                                                  VNC 1 S VMIX1
                                                                                          ECF AR, ARAT
                                          SATI, PMIX1
                                                                                                     EL1, VEL2
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SECTOR LB
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PAL = SAME AS PSAT
PAL = LOWEST SATURATION PRESSURE THE STEA
CORRESPONDING SATURATION TEMPERATURE FALL
COCLANT TEMPERATURE. THE RESULT IS O HEAT
TAL = SAT STEAM TEMP IN RANKIN
STSAT = STAT STEAM TEMP F
VG = SPECIFIC VOLUME OF STEAM FT**3/LB
AMMINC = SPECIFIC VOLUME OF STEAM AND N/C GASSES
VNC = SPECIFIC VOLUME OF STEAM AND N/C GASSES
VNIX = SPECIFIC VOLUME OF STEAM AND N/C GASSES
WMS = STEAM FLOW TO A RUW OF TUBES IN A SECTOR
WGAS = FLOW OF N/C TO A SECTOR LB/HR
AOTFLW = AR EA OPEN TO FLOW IN A BANK OF TUBE
VEL = STEAM-N/C MIXTURE VELOCITY IN A ROW O
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HETTRN
                                                                                   PAL=PSAT(1, J)
TAL=TSATFN(PAL)
STSAT(1, J) = TAL-459.69
VG = VGFN(TAL, PM IX(1, J))
VN C=(10.729*TAL)/(AMWNC*PM IX(1, J))
VN IX=1.0/((AMDLST)/(AMMNC*PM IX(1, J))
VEL(1, J)=(WS(1, J)+WGAS(J))*VM IX/(3600.0*ADTFLW(1))
AMOLSC=WNC/AMWNC
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PRICR TO CALLING
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MAIN ROW LOOP
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                                                                                                                                                                                                                                                                                                                                                                                                                              I = ROW NUMBER DUMMY VARIABLE AXO = AREA OPEN TO FLOW IN A ROW AOTFLW = AME AS AXO L = SECTOR NUMBER DUMMY VARIABLE J = SECTOR NUMBER OF ROWS OF TUBES DELODF = CHANGE IN TUBE OD / ROW
                                                                                                                                                                                                                                                                                                                                                                                    THIS SECTION IS INSIDE THE INITIAL IZES SOME VARIABLES IN SEC-5.
NR WSP3=NDROWS+1
DE LPP=PTLIM/(RTG+1.)
CHECK2=-2.
DO 80 M=1,NRWSP3
PS AT(M, J)=PTLIM-(M-1)*DELPP
CONTINUE
PM IX(1, J)=PSAT(1,J)/(AMOLSS/AMOLST.
                                                                                                                                                                                                      SECTION
                                                                                                                                                                                                                                             SE CA3=0.

IF (SECA3.NE.1.) GD TD 100

WR ITE (6,770) PMIX(1,J),PSAT

WR ITE (6,780) STSAT(1,J),TAL

WR ITE (6,790) VG,VNC,VMIX

WR ITE (6,800) VEL(1,J)
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FACTOR ENHANCEMENT
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                                                                                                                          STSAT = STEAM SATURATION TEMP F
ANT = TARGET TUBE LOCATION IN VERTICAL REWARS
WNS = STEAM FLOW TO A ROW OF A SECTOR LEBM/HR
UN = OVERAL L HEAT TRANSFER COEFFOR FOR ENDMONT
ALM TO = LMTD FOR A ROW IN A SECTOR
HOMCI = STSAT - COOLANT OUTET TEMP
STFO = AVE TEMP OF OUTER TUBE FILM
STFO = AVE TEMP OF OUTER TO BE FILM
STRAIL TO F N/C GAS TO STM AT ONE ROW OF TUBES
SH = INTERNAL HEAT TRANSFER COEFF
ROUT = EXTERNAL HEAT TRANSFER COEFF
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HOMCO=0.

VP SHH(1, J)=1.E+8

HE FF(1, J)=1.E+8

RC(1, J)=1.61E-3

VN RE(1, J)=1.61E-3

VN RE(1, J)=1.65E+5

SHI(1, J)=1.36E+5

SHI(1, J)=1.36E+5
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IF IT HAS CHECKISO AND DUMMY VALUES HAVE ALREADY BEEN ASSIGNED
TO WCND.
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     BELOW
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CONTINUE
JX=0
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 IF CHECK2 < 0. THAT INDICATES THAT THE SATURATISE BELOW PILIM AND THUS THE STEAM TEMPERATURE INLET TEMPERATURE. THIS MEANS THAT NO HEAT IS STEAM AND THE ALMTD IS 0. THE COOLANT OUTLET TECOLANT INLET TEMP AND THE TUBE FILM TEMP IS THE COOLANT INLET TEMP. THE REMAINING VARIABLES ARE GOOD VALUES RETURNED BY HETTRN.
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  WCND(I, J)=UN(I, J)*AO*ALMID(I, J)/HFG*IBNPR(I)
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AND DUMMY VARIABLES HAVE BEEN A
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HF G=HF GFN(STSA T(I,J))
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HO MCI = 0.
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ALLOW THE PROGRAM
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(6,1020) I,J
(6,970) WS(I,J),WS(I+1,J)
(6,1000) PSA†(I,J),PSA†(I+1,J)
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+1, J) *AMOL S/ (AMOL S+WNC/ AMWNC
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MIX(I+1, J)=.001
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PAL=PSAT(I+1, J)
IF (I .EQ.I) DELPTP
PMIX(I+1, J) = PMIX(I
IF (PMIX(I+1, J) - LE
GO TO 230
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                    S(I+1,1).LE.
S(I+1,1).LE.
                                    COMPUTING
                                                                                                                                 PM IX(I+I+O)
PSAT(I+I+O)
PAL=PSAT(I+I+O)
PTEST=PSAT(I+I+O)
IF (SECA6-NE-I+O)
IF (SECA6-ED-I)
IF (SECA6-ED-I)
                                                                                                                                                                                                                          IK 2=1+1
SE CA6=0
NR WSP 2=NOROWS+1
RO TOG=FLOAT(NR W
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                                             ALCULATE PTST WHICH IS A TEST VALUE INDICATING THAT SATURES SLRES WERE REACHED IN THE ANALYSIS FROM WHICH THE ORRESPONDING TSAT IS EXCEEDED BY THE INLET COOLANT TEMP() AS VIOLATED! A LARGE NEGATIVE VALUE FOR PTST INDICATES A ONDENSOR DESIGN IN WHICH PTLIM WAS VIOLATED EARLY IN THE NALYSIS. AS THE CONDENSOR MODEL IMPROVES AND SATURATION OF STAYS ABOVE PTLIM THEN PTST CONVERGES TO 0.
                                                                                                                                PSAT(I, J), PSAT(I+1, PTST(J)
                                                                                                                                                                                                    OF TUBES
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| X(I, J) - DELP TP
| X(I, J) = 001
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                                                                                            PT ST (J) = (X1/(X2-X1))-FLOAT (NOROWS-I)
CHECK2 = -2.
PA L=P SAT(I+1, J)
PM IX(I+1, J)=PM IX(I, J)-DELP TP
IF (PM IX(I+1, J)-LE..001) PM IX(I+1, J)=
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COEF FOR
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                  JI-DELPI
                                                                                                                                                                                       PAL = SAT STM PRESS
STFO = AVE TEMP OF OUTER
UN = OVERAL HEAT X-FER
AO = TUBE OUTER SURFACE
ALM TO = LMTD
TBNPR = NUMBER OF TUBES
WS = STEAM FLOW TO A ROW
WNC = N.C. FLOW TO A ROW
JGAS = FLAG FOR TYPE OF
AMWNC = MOLECULAR WEIGHT
TAL = SAT STM TEMP ENTER
WCND = CONDENSATE FROM
UCND = CONDENSATE FROM
OUTER
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            NRMS P2
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            DO 220 IK=1K2, N
PSAT(IK, J)=PSAT
CONTINUE
XI=PTEST-PTLIM
X2=PSAT(I, J)
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TAL-TSAT) ) / HFG | .J) *CPS FN(TSAT) / 18.015+1 | WCND(II, J), WCNDP
                                                                                                                                                                                                   CHECK TO SEE IF WCND IS O WHICH CAN OCCUR IF PSAT HAS DRUPPED BELOW PTLIM (CHECK2<0). IF IT HAS THERE IS NO HEAT TRANSFER FROM THE STEAM AND NO NEED TO CORRECT FOR SENSIBLE HEAT
                                                                                                                                                                                                                                                                                                                                         WC ND
                                                                                                                                                                                                                                                                                                                                        10
                                                                                                                                                                                                                                                        CHECK TO SEE IF STEAM FLOW HAS ALREADY GONE TO O (CHECK<0)
AND WCND HAS BEEN FIXED UP WITH DUMMY VALUES
                                                                                                                                                                                                                                                                                                                                       EITHER CHECK IS TRUE THEN AVOID CORRECTIVE ITERATION
                                                                                                                                                                                                                                                                                                     AVOID ITERATIONS
AMOLS = MOLES OF STEAM IN A ROW OF A SECTOR PAIX = PRESSURE OF STEAM - N/C MIXTURE PSAT = SATURATED STEAM PRESS TSAT = SATURATED STEAM TEMP R STSAT = SATURATED STEAM TEMP F
                                                                                                                                                                                                                                                                                                                                                          290,260,260
                                                                                                                                                                                                                                                                                                                                                    IF (ABS(WCNDP/WCND(I,J)-1.0)-.0051 290,260,24

IF (JX-50) 280 280 270

IF (SECA7.EQ.I) WRITE (6,1120) IDENT.I.J.I.J.

GO TO 250

WCND(IJ)=WCNDP

GONTINUE

SECA6=0.
                                                                                          141, 11-51F01 240,240,250
                                                                                                                                                                                                                                                                                                      0
                                                                               TSAT= TSATEN(PAL)
SECA7=0.
ST SAT(1+1, 1)=TSAT-459.69
IF (ST SAT(1+1, 1)=STF0) 240.240.2
ST SAT(1+1, 1)=STF0
TSAT=STSAT(1+1, 1)+459.69
CONTINUE
WCNDP=(UN(1, 1) *AO*ALMTE(1, 1)*TBN
1WNC*CPAFN(TSAT, JGAS)/AMWNC)*(TAL
IF (SECA7-EC-1) WRITE (6,1070) WIF (SECA7-EC-1)
                                                                                                                                                                                                                                       IF (CHECK2.LT.0) WCNDP=WCND(I,J)
                                                                                                                                                                                                                                                                                    (CHECKI.LT.0) WCNDP=WCND(I.J.
                                                                                                                                                                                                                                                                                                     CONDENSATE FLOW HAS GONE TO
                                                                                                                                                                                                                                                                                                                       (WCND(I, J). EQ.O.) GO TO 290
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                                                                                   ALLOW THE PROGRAM TO CONTINUE AND DUMMY VALUES FOR STEAM FLOW AND CONDENSATE ARE ENTERED INTO THE REMAINING ROWS. IF STEAM FLOW HAS GONE TO I THAT INDICATES ZERO STEAM FLOW. A VERY SMALL NUMBE WILL BE USED TO SIMULATE O STEAM FLOW IN THE REMAINING ROWS.
      ROW
DUMMY
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WITH
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      EEN FIXED UP
XUP HAS BEEN
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F ROWS HAVE BE
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                                                                                                                    IK 1=1+1
ROTOGO=FLOAT(NDROWS-I)
WS END=(WTEST-1.1/WS(I.J)-WTEST+1.)
WT ST(J)=WSEND-ROTOGO
                                                              STEAM
                                                                                                                                                                       OR
                                                              IF
                                                                                                                                                                       FLOW RAT
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TUBDRY
I,J)/TBNPR(I)
SUBRAT
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ST060+11
      E IF STEAM FLOW I AND SUBSEQUENT EGATIVE CHECKI
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(AST 060+1)
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BDRY=TUBDRY+TBL
BDRY=TUBDRY-WSL
JBDRY=CUMDRY+TU
UBRAT=KCND(I-1;
UBAVE=SUBAVE+SU
BAD=NBAD+1
                                                                                                                                                                                                                                          R WSP1 = NOROWS+1
S TOGO = FLOAT(NRE
S (1, 1) = (1, 0) * (R.
CND(1, 1) = WS(1, 1)
ELWS = WS(1, 0) / (R.
                                                  = WS(I+1,J)
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CALCULATE A VALUE FOR THE TEST VARIABLE (WTST) INDICATING O DK
NEGATIVE STEAM FLOW. A LARGE NEGATIVE VALUE FOR WTST INDICATES
A VERY EAD CONDENSOR MODEL WHERE NEGATIVE STEAM FLOW WAS
ENCOUNTERED EARLY IN THE SECTORS. AS THE CONDENSOR MODEL IMPROVES
STEAM FLOW STAYS POSITIVE THROUGHOUT THE CONDENSOR AND WTST
CONVERGES TO 0.
                                                                                                                                                                                                                                                                              580
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900.
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                                                                                                                             PRESSURE ARRAY (CHECK2<0)
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                                                                                                                                                                                                                                                                              I E
                                                                                                                                                                                                                                                                  SEC TOR
                                                                                                                                                           +1, J}=PM IX(I +1, J)*AMOL S/ (AMUL S+WNC/AMMNC SATFN (PS AT (I +1, J))
I+1, J)=T SAT-459.69
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OMPLETED
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                                                                                                                                                                                                                                  . 51 SAT(I, J)
(I, J), PSAT(I, J)
                                                                                                                                                                                             SEC TION
                                                                                                                                                                                                                                                                  MAIN
DO 330 IK=IKI, NRWSP1
IF (SECA6.EQ.1) WRITE (6,970)
WS(IK,J)=WS(IK-1,J)-DELWS
WCND(IK,J)=WCND(IK-1,J)-DELWC
                                                                                                                                                                                                                                                                              SECTOR
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FIC VOLUME
                                                                          CHECKI = -2.

IF (I I I I = Eq.1) GD TD 180
GD TD 350
WCND(I + J) = WCND P
WS (I + I + J) = WS(I + J) - WCND (I + J)
GFL W(I + J) = (WS(I + J) + WNC) / AX
AMOL S = h S(I + I + J) / 18 * 015
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AND DUMMY VALUES ARE IN
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CONDENSER
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D A SECTOR

EAM FLOW IN A ROW OF TUBES
FLOW TO MATCH PRESS DROPS IN SEC-10
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   VG = VGFN(TSAT,PMIX(I+1,J))
AMOLS C=WNCAAMMNC
VNC=(IO.729*TSAT)/(AMWNC*PMIX(I+1,J))
VNC=(IO.729*TSAT)/(AMMNC*PMIX(I+1,J))
VMIX=IO.0/((AMDLS/(VG*(AMOLS+AMOLSC)))+(AMOLSC/(VNC*(AMOLS+AMOLSC))
AMOLS = MOLES OF STM TO A ROW
PISAT = STM SATURATION TEMP R
PMISAT = STM SATURATION TEMP R
PMISAT = STM SATURATION TEMP R
PMISAT = STM SATURATION TEMP R
VNC = SPECIFICAL ANC DIVINE OF N/C GASSES
VNC = SPECIFICAL ANC TO A SECTOR
AMOLES OF A N/C TO A SECTOR
VNC = STM CF COW TO A SECTOR
AMOLES OF A N/C TO A SECTOR
NS = STM FLOW TO A ROW IN A SCOUND OF TOBES
NS = STM FLOW TO A ROW IN A SCOUND OF TOBES
NS = STM FLOW TO NET TO STM FLOW TO B TOBE IN THE CONDENSER
NS = STM FLOW TO NET TOBE IN THE CONDENSER
NS = SUM OF COLLANT TO UTLET TEMP FOR A TUBE
SM B = ROW BY ROW SUM OF COLLANT TO UTLET TEMP FLOW RATE)
SM B = ROW BY ROW SUM OF COLLANT TO UTLET TEMP FLOW RATE)
SM B = ROW BY ROW SUM OF COLLANT TO UTLET TEMP FLOW RATE)
SM B = ROW BY ROW SUM OF COLLANT TO UTLET TEMP FLOW RATE)
SM B = ROW BY ROW SUM OF COLLANT THROUGH THE CONDENSE
ND = OVERALL HEAT TO UTLE R SURFACE AREA
ALM TO = ACCOUNT TO UTLET TEMP FROM INLET TO CURRENT ROW
ALM TO = ACCOUNT TO DRIVE THE COLLANT THROUGH THE CONDENSE
PHP = ACCOUNT THE COLLER TO BREAD ACROSS A ROW
TALL SAT STEAM FLOW FROM TO STEAM FLOW TO NIXTURE ENTERING CONDENSE
SW IT = DUMMY LOOP VERLE IT = 1 / R ROW OF TO BE TO RESE OF STM TO NIX E SOUR ENTER TO NIXTURE ENTERING CONDENSE
NOT A STANDARD SECTOR
SW SS IN E ROW OF TO STEAM FLOW TO MATCH PRESS DRORPS IN SECTION
MORS = FLOW OF NIC GAS TO A SECTOR
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|+1,J)+WNC)/(AOTFLW(I+1))*VMIX/3500.0
|TBI*TBNPR(I)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                J) *TBNPR( I
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0*ALMTD(I
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VEL(I+I, J) = (WS (I+I, J) +
SM TBI = SM TBI + WB * STBI * TE
SM WB = SM KB + WB * TBN PR (I)
SM TBZ = SM TBZ + WB * TBN PR (I)
SU MQ = SUMQ + UN (I, J) * AO*
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SURE VALUES IF
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INDICATING STEAM
N ADDED TO EXITER
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                                                                                                            HAS NOT
                                                                                                                                                          (CHECK1.GE.O) STMSUM=STMSUM+WS(NOROWS+1,
                                                                                                                                                                                                                                                                                                                                                                                                                   ADJUST
                                                                                                           E STEAM FLOW
DUMMY VALUES
                                                                                                                                                                                                                                                                                                                                                                                                                   PRESS
QO A(I, J)=UN(I, J)*ALMTD(I, J)
CUMDP(I, J)=(PMIXI)-PMIX(I+1, J)
DELP(J)=DELP(J)+DELPTP
TAL=TSAT
CONTINCE
C(J)=(WS(I, J)+WGAS(J))/SQRT(DELP(J))
                                                                                                                                                                                                                                       EMP DUTPUT SECTION
                                                                                                                                                                                                                                                                                                                                                                                                                                                            PM XEX T=0.

DE LPVE=0.

PS UM=0.

VE LEX T=0.

VE LEX T=0.

DO 420 I=1, ISE C

DO 420 I=1, ISE C

PA L=P S AT (K S T O P

PS UM=P S UM + PAL/ S ECF L G

VE LEX T= VELEX T+ VEL( NOR O W S + 1, 1)

DE LPVE= DELPVE + DELP ( I )

PM XEX T= PM X EX T + PM IX (NOR O W S + 1, 1)
                                                                                                                                                                                                                                                                                                                                                                                                                    AND
                                                                                                           CNLY THOSE SECTORS WHER UM SO AS NOT TO INCLUDE STEAM FRACTION
                                                                                                                                                                                                                                                                                                                                                    WTST(II), PTST(II)
                                                                                                                                                                                                                                                                                                                                                                                                                  EXIT FRACTION
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THIS IS DONE
THE VARIABLE
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TST OF ALL
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MAKE IT CONTINOUS FROM 1-->-1 REFLECTING THE CONDENSOR IN STEAM IS EXTRACTED AND THE CONDENSOR WHERE ALL THE TUBES AR
                                                                                                         PTST(I)=PSAT(NGROWS+1, I)-PTLIM
                                                                                                                                                                                                                                  IF (ABS(DELPVE).GT.00.01) 450,450,460

IF (ABS(DELPVE).GT.00.001) GO TO 460

CONTINUE

                                                                                     DO 430 I=11SEC
IF (PTST(I).GE.O.) PTS
WT SAV=WTSAV+WTST(I)
FR=-(SUBFLO)/WSI
DELPVE=CELPVE/SECFLG
DO 450 J=1, ISEC
     RY SA
                                                                                                                                                                                                                                                                      440
450
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WP T=WP T+WP(J) CONTINUE WS P(JMA X) = W SI - WS PT WS P(JMA X) = W SI - WS PT WS P(JMA X) = W SP (JMA X) WG AS(JMA X) = W SP(JMA X) * WNCFR WP (JMA X) = W SP(JMA X) + WG AS(JMA X) IP (LOOP = IP (LOCP + 1)	CHECK FOR STEAMM ADJUSTMENTS WHICH TRY TO PUT TO LOW VALUE INTO A SECTOR		00 T0 550 MSP(IFAIL)=2.*COMP-WSP(IFAIL) MSP(IFAIL)=2.*COMP ADJ=DIFF/FLOAT(ISEC-I) DO 540 J=1 FISEC IF (J.EQ.IFAIL) GO TO 540 WSP(J)=WSP(J)-ADJ		INCORPORATE FR INTO EXITER TO MAKE IT CONTINUOUS FRUM	EXITER=STMSUM/WSI+FR WSEXIT=STMSUM/WSI+FR WSEXIT=STMSUM/WSI+FR VELEX T=VELEXT/SECFLG DELPVE = DELPVE/SECFLG PM XEX T=PMXEXT/SECFLG PM XEX T=PMXEXT/SECFLG TS ATEX=TSATFN(PSUM) AM LSEX=STMSUM/18.015 IF (STMSUM-LE.0.) AND.(PRCCLR.GT.0.) AMLSEX=I.0/16.01 IF (FCCLR.EQ.0.0) EXITDA=EXITFR
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IF (PRCCLR.EG.O.O) GD TO 590 CALL TO COOLER, IF THERE IS ONE ENHICE ENHIN ENHICE ENHIN TABLE TO SHAIN TABLE TO
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AVIS = AVG VISCOSITY OF STEAM-N/C GAS MIXTURE
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SEE EQUATION 11 IN APP B
DD = DIFFUSIVITY (SEE NOTE)
DDG = DUMMY PASSING PARAMETER FCR SUB DS FVTY
E20K = N/C GAS FORCE CONSTANT (SEE SUB DS FVTY)
PATM = SAT PRESS IN ATM
PGB = NONCOND PARTIAL PRESSURE
PSAT = STEAM SATURATION PRESS PSI
R = N/C GAS COLLISION DIAMETER (SEE SUB CSFVTY)
TSAT = STEAM SATURATION TEMP IN DEGREES KELVIN
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FER COEF BTU/HR-FT**2-HE COCLANT BTU-FT/HR-F
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BB = COOLANT FLOW IN TUBE LBM/HR-FT UBE

BMU = COCLANT VISCOSITY LBM/HR-FT

BNF = TUBE FLOODING FACTOR

CBI = COCLANT VISCOSITY LBM/HR-FT

BNF = TUBE FLOODING FACTOR

CBI = COCLANT CONCENTRATION

CBI = COCLANT CONCENTRATION

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SHBI = SPECIFIC HEAT TRANSFER COEF

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FDAVE = INPUT VALUE OF TUBE FLOODING FACTOR (SEE NOTE BLOW)
FDAVN & FDAVM ARE INTERMEDIATE VALUES USED LATER.
FOUL = TUBE FOULING FACTOR INPUT BY USER
RFACT = SUM OF THERMAL RESISTANCES NECLECTING GAS FILM
RIN = RESESTANCE TO HEAT X-FER DUE TO INTERNAL FILM
SHNF = EXTERIOR HEAT X-FER COEF, SHMK MODIFIED FOR RAIN
SHWINV = INVERSE OF WALL HEAT TRANSFER COEF
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IF (FDAVE.LT.O. GR.BNF.LT.2.) GO TO 160

FDAVN=0.6*FDAVE+(1.-0.5647*FDAVE)/BNF**.20

FDAVN=0.6*FDAVE+(1.-0.5647*FDAVE)/(BNF-1.)**.20

SHNF=SHMK*(BNF*FDAVN-(BNF-1.)*FDAVM)

GO TO 170

SHNF=0.95*(BNF**0.9-(BNF-1.)**0.9)*SHMK

RF ACT=RIN+SHMINV+FOUL+1./SHNF

IF (HETCN.NE.1.) GO TO 180

WR ITE (6,310) M5

WR ITE (6,540) SHNF.RFACT
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RODT = RATIO OF HOMCI TO HOMCO
STBI = COOLANT INLET TEMP
STSAT = STEAM SATURATION TEMP
TB2 = COCLANT OUTLET TEMP
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IF (RODT-6T-1-1) GO TO 190
AL MTD=0.5*(HOM CI+HOMCO)
IF (HETCN-EQ-1-) WRITE (6.670)
GO TO 200
AL MTD= (HOMCI-HOMCO)/ALOG(RODT)
CONTINUE
IF (HETCN-NE-1-) GO TO 210
MS = 10
WRITE (6.310) MS
WRITE (6.550) HOMCI-HOMCO
IF (WNC-NE-0-) GO TO 230
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DGFLM = TEMP DROP ACROSS GAS FILM, SEI
DLFLM = TEMP DROP ACROSS LIQUID CONDEN

K = COUNTER FOR NUMBER OF ITERATIONS

RFACT = SUM GF THERMAL RESISTANCES

ROUT = EXTERNAL FILM HEAT X—FER COEF

SHNF = EXTERNAL FILM HEAT X—FER COEF

SKBO = THERMAL CONDUCTIVITY OF EXTERNA

ST CO = TEMP AT SUFFACE OF OUTER LIQUID

ST SAVE AVE COOLANT TEMP

ST SAT = STEAM SATURATION TEMP

ST SAVED VALUE OF UEST FROM LAST ITERATI

UESTS = SAVED VALUE OF UEST, USED ONLY

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LM = TEMP DIFF ACROSS CONDENSA

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SFLM = DGFLM FROM PREVIOS ITERA

F = N/C GAS HEAT X-FER COEF

F = N/C GAS HEAT X-FER COEF

F = N/C GAS HEAT TOF VAPOR

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STCO= STSAT

STFO= (STCO+STWO)/2.

DG FLM= 0.

DL FLM= STCO-STWO

DELFLM=DLFLM

K= K+1

IF (HETCN.NE.1.) GO

MS = 11

WR ITE (6,310) M5

WR ITE (6,540) WTEST

WR ITE (6,590) STWO.

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RFACT = SUM CF THERMAL RESISTANCES NEGLECTING N/C GASSES

ROUT = EXTERNAL HEAT X-FER COEF

DD = TUBE DD

SHNF = EXTERNAL LIQUID FILM HEAT X-FER COEF

SKBO = THERMAL CONDUCTIVITY OF OUTER LIQUID

SKBO = THERMAL CONDUCTIVITY OF OUTER LIQUID

ST BAVE = AVG COOLANT TEMP

ST CO = TEMP AT SURFACE OF OUTER LIQUID FILM

ST C = DUTER LIQUID FILM AVG TEMP

ST ST = TUBE OUTER WALL TEMP

T = AVG LIQUID FILM TEMP IN RANKIN

TO IF = TEMP OIF BETWEEN SAT STM AND AVE COOLANT TEMP

TSAT = STM SAT TEMP IN RANKIN

UEST & UESTS V - VALUE OF UTEST AND UEST FROM PREVIOS ITER

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TB2 = TUBE OUTLET
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MMON /INPT/ BC(6) BE(6) ENH(6), ENI(6), ENO(6) FOUL, PHI, SKW, SAII, TUBESW, WNCIR, IOPI, PTST(15), SECWID, IBAF, ISEC, JBAF(16), JGM, MPITCH, NEE, NEI, NETE(6), NETI(6), NRNE(6), NRNI(6), NUCROWS, WTSI, PFILL

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UBE X-SECTIONAL OCLER **21/62 1-DELPCI-(VELC(1)**2)/62 (1)*AMLSEX/(AMLSEX+AMLSCC) щш OH. 1) 0,40,30 A2DA1)*VELC(1)**2)/G; (THE RATIO VL CMAX = VELC(1) G2 = 288 • 0*56*SVMXEX IF (VELC(1) - VELEXT 10 • 10 • 20 A2 DAI = VELC(1) / VELEXT DE LPCT = ((1.0-A2DAI)*VELEXT**2) / G2 G0 T0 50 A2 DAI = VELEXT/VELC(1) IF (A2 DAI - 0.715) 40,40,30 DE LPCT = (0.75*(1.0-A2DAI)*VELC(1)**2 G0 T0 50 DE LPCT = (0.4*(1.25-A2DAI)*VELC(1)**2 PM IXC(1) = PMXEX T-DELPCT - (VELC(1)**2-PMXEX T-DELPCT - (VELC(1)**4-PMXEX T-DELPCT - (VELC(10 RATIO AREA) T d Ø IE AREA TTUBE 20

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         SECFLG=FLOAT(ISEC)
TS ARE A=PI*(BNDRAD**2)*SECFLG*SECWID/360.
AHOLE S=ARATIO*TSAREA
AHOLES=AHOLES+(PI*SDO**2)*TNOC/4.
TS ARE A=TSAREA+HTCLR*HNOC*S CDC*SDOC
AR ATIO=AHOLES/TSAREA
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                                                                                                                        IF (PSATC(1).GT.PTLIM) GO TO 8C SUMOC=0.

NR WSP 4=IVNOC+1

CONSI B=PMXEXT/FLOAT(NRWSP4)

CONSI B=PMXEXT/FLOAT(NRWSP4)

CUMDP C(1)=PMIXI-PMXEXT

PS ATC(1)=PMIXI-PMXEXT

OO AC(1)=PMXEXT

OO AC(1)=0.

DE LPC=PMXEXT

WC(1)=WSEXIT

WC NDC(1)=0.
                                                                                  VOLC=HTCLR*SDDC*SDOC*HNOC*ALST
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SIA
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UNC(M)=700.
QO AC(M)=0.
AL MTDC(M)=0.
CONTINUE
SM WBC=SMWBC+WB*TNOC
SM TBIC=NB*TNOC*STBI
SM TBZC=SMTBIC
EXITFC=1.
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WS C(M) = WSEXIT
WCNDC(M)=0.
PM IXC(M)=PMIXC(M-1
PS ATC(M)=PSATC(M-1
CUMDPC(M)=PMIXI-PP
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IF (FDAVE-LT.O.-OR-BNF.LT.2.) GO TO 150

FDAVN=0.6*FDAVE+(1.-0.5647*FDAVE)/BNF**.

FDAVM=0.6*FDAVE+(1.-0.5647*FDAVE)/(BNF-SHMK*(BNF*FDAVE)/(BNF-1.)*FDAVE)/(BNF-1.0.160)

GO TO 160

SHNF=0.95*(BNF**0.9-(BNF-1.)**0.9)*SHMK
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IF (TERM•LT••278) GD TO 130
AN USP = 1.24*(ANUS**0.8)*(RETP**0.1)
GD TO 140
AN USP = ANUS.*0.70/0.725
SH MKP = ANUSP*SK BC/S DO
SH MK = SH MKP
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60 TO 49
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                   SHMK1=SHMK
AN US=SHMK*SDD/SKBD
EMUL=BMUFN(0., STSAT)
EMUS=SMUFN(TSAT)*3600.
RHOL=RCEFN(0., STSAT)
RHOS=1./VGFN(TSAT, PSAT)
VISRAT=(EMUS/RHOS)/(EMUL/RHOL)
RETP=XNRE*VISRAT
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AND CHANGE 0
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                                                                                    HOMC I, HIMCO, LJ, IR
                                                                                    5301
                                                                                    WRITE (6.
HIMCO=1.E
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                                                                                    (HIMCO.LT.1.E-1
(HIMCO.LT.1.E-1
                                                                                                                                                  RODT = HCMC I / HIM CO
                        HO MCI = STSAT-ST B
HI MCO= STSAT-TB 2
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THE HF G=HF GFN(STSATC(I))
WCNDC(I)=UNC(I)*AO*HNOC*ALMTDC(I)/HSTO
CALL PRSDRP (TAL,VMIX,WSC(I),WNCC,AXDC,SDO,VPSHC(I),DELTPC,ENHF)
JY = 0
WSC(I+1)=WSC(I)-WCNDC(I)
GFLOWC(I)=(WSC(I)+WNCC)/AXOC CALCULATING EATE A TEST VARIABLE INDICATING POINT IN COOLER WHERE STEAM D AND A NEGATIVE FLOW RATE FOR USE IN CALCULATING EXITFRC AFTER NEG IF STEAM FLOW HAS ALREADY GONE NEGATIVE THEN AVOID TEST VARIABLE 9 IF PREVIOUS COND. RATE PREDICTS STEAM FLOW WILL NEXT ROW, SET CONDENSATE EQUAL TO STEAM FLOW. NR DDR Y = I VNOC-I TB DRY C = TNOC * (FLOAT (NR DDRY) / FLOAT (I VNOC)) TB DRY C = TBDR YC - WTST C * HNOC SB RAT C = WCNDC(I) / HNOC SB FLOC = SBRATC* TBDR YC STC= (WSC(I+1)-1.)/(WSC(I)-WSC(I+1)+1 STC=WTSTC-FLOAT(IVNOC-I) OWC=1 COMPLETICN OF TUBE H/T CALCULATION CHECK FOR NEGATIVE STEAM FLOW (WSC(I+1)-1) 300,300,360 (IRGMC.GT.0) GD TO 310 RO UT=1 - / ROUT UN C(I) = VNF AL MTDC(I) = ALMT D SH IC(I) = SHI SHNFC(I) = SHNF RC C(I) = RC RO UTC(I) = RC VN REC(I) = XNRE VP SHHC(I) = VPSHH HE FFC(I) = SHEFF WC NDC (I) = MSC (I I.F I.F. BAB 77 0 0 0 0 SOU $\circ\circ\circ$ S S 0000



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STSATC(1+1)=SFFC PM NESTFACE (1+1)=SFFC PM NESTFACE (1+1)=1. ENTER DUMMY VALUE FOR NEXT ROW WSC ENTER DUMMY VALUE FOR NEXT ROW WSC WS C(1+1)=1. GGC=(WSC(1)*CP SFN(TAL)/18.015+WNCC*CPAFN(TAL,JGAS)/AMWNC)*(TAL-TSA TG STSATC(1)=1. GGC=(WSC(1)*CP SFN(TAL)/18.015+WNCC*CPAFN(TAL,JGAS)/AMWNC)*(TAL-TSA TG STSATC(1)=TB2) TG STSATC(1)=T
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                                                                                                                                                                                                                                                                                                                                                                                                      IXI-PMIXC(IFX-WB*HNOC*STBI
                                                                                                                                                                     出上
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                        LPC = PMIXC(1) - PMIXC(NRWSP4)
                                                                                                                                                                     ATE PARAMETERS FOR
                                                                                                                                                                                                  DO 370 M=NRWSP5 .NRWSP4
QOAC(M)=0.
WS C(M)=WSC(M-1)
WC NDC(M)=0.
ST SATC(M)=STSATC(M-1)
PS ATC(M)=PSATC(M-1)-CON
PM IXC(M)=PENIXC(M-1)-CON
VELC(M)=VELC(M-1)
  SMWBO+WB * HNOC
= SMTBIC+WB*HNOC*
= SMTB2C+ WB*HNOC*
                                                                                                                                                                                                                                                                                                                                                                    DO 380 IFX=NRW SP51 IV
UNC(IFX)=UNC(IFX-I)
CUMDPC(IFX)=PMIXI-PM)
SM TBIC=SMTBIC+WB*HNOC
SM TB2C=SMTBIC
OO AC(IFX)=0
SM NBC=SPWBC+WB*HNOC
ALMIDC(IFX)=0
                                                                 NR WSP4=IVNOC+1
RTG=FLOAT(NRWSP4-I)
NR WSP5=I+1
CONST=PMIXC(I)/RTG
CONST2=PTLIM/RTG
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                         ,290
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GT 00)
(6,550)
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   WBC=
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AMWNC*PMIXC(I+1))
(VG*(AMLSSC+AMLSCC)))+(AMLSCC/(VNC*(AMLSSC+AMLSC)
   WR ITE (6,540) I
CONTINUE
WCNDC(I)=WCNDP
WS C(I+1)=WSC(I) -WCNDC(I)
AM LSSC=WSC(I+1)/18.015
PS ATC(I+1)=(PM IXC(I+1)*AMLSSC)/(AMLSSC+AMLSC)
TS ATC(I+1)=(PM IXC(I+1)*AMLSC)/(AMLSSC+AMLSC)
TS ATC(I+1)=TSATC-459.69
ST SATC(I+1)=TSATC-459.69
VG=VGFN(TSATC,PMIXC(I+1))
VN C=10.729*TSATC/(AMWNC*PMIXC(I+1))
VN C=10.729*TSATC/(AMWNC*PMIXC(I+1))
                                                                                                                                                                                                                                                                                                 VELC(1+1)=(WSC(1+1)+WNCC)*VMIX/(AXOC*36C0.0)
VELC(1+1)=(WSC(1+1)+WNCC)*VMIX/(AXOC*36C0.0)
SM TBIC=SMTBIC+WB*HNOC*TBI
SM WBC=SMTBSC+WB*HNOC*TBI
SUMQC=SUMQC+UNC(I)*AO*ALMTDC(I)*HNOC
QO AC(I)=UNC(I)*ALM TDC(I)
TAL=TSATC
AN FC=ANFC-1.0
CONTINUE
DELPC=PMIXC(I)-PMIXC(IVNOC+1)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           NEGATIVE
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EXITEC=-(SBFLOC)/WSC(1)
RETURN
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 480
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           IF (IRDWC.LT.2) GO TO 4

NR WSP4=IVNDC+1

SUMQC=0.

DO 470 M=1.NRW SP4

GO AC(M)=0.

AL MTDC (M)=0.

WC (M)=0.
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                      EXITFC= NSC(I+1)/WSC(1)
WSOUT=WSC(I+1)
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CONSECUTION OF THE CONVERGENCE IN FILM TEMPERATURE S, 84 DLUFLM= FIZ CONSESSOON OF 22 H NO CONVERGENCE IN USEY. UEST = 'FIZ-5.94 UTEST = 'CONSESSOON OF 22 H NO CONVERGENCE IN DELFLM, DELFLM= FIZ-5.94 UTEST = 'CONSESSOON OF 23 H NO CONVERGENCE IN DELFLM= FIZ-5.94 UTEST = 'CONSESSOON OF 23 H NO CONVERGENCE IN DELFLM= FIZ-5.94 UTEST = 'CONSESSOON OF 23 H NO CONVERGENCE OF CONSESSOON OF 24 H NO CONVERGENCE OF CONSESSOON OF 25 H NO CONVERGENCE OF CONSESSOON OF CONSESS
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1 (1H0,53H SAT STEAM TEMP LESS

2-5,10H STBAVE = ,E12-5,9H SE

1 (1H0,25HNO CONVERGENCE FOR WC

(1H0,68H**** SATURATED STEAM

PMIXC = ,E15-5,6H ROW,15)

(1X,52HERROR,COOLER PTLIM VIOL
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2.0. SHIP CELL OF LOAT (100) . WCNDC(100) . GFLOWC(100) . GGR36 627

2.0. SHIP CELL OF LOAT (100) . WCC(100) . WCNDC(100) . GFLOWC(100) . CG033 627

1. BCCARN . CCDRAC . ROAT (100) . WCC(100) . WCNC(100) . GGR37 CD0

1. BCCARN . CCDRAC . SHIP CCDRAC . SH
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DT CND 2= (AVTB2-AVTB1)/ALOG((STSAT1-AVTBI)/(STSAT1-AVTB2))
UP CGN D=SUMQ/(AOT*DTCND2)
WR ITE (6,80)
WR ITE (6,90)
WR ITE (6,90)
WR ITE (6,90)
WR ITE (6,90)
WR ITE (6,100)
WR ITE (6,100)
WR ITE (6,100)
WR ITE (6,110) UBAR W, ADTCND, DELPVE, TOROP 1, VEL2, VEL3
WR ITE (6,120) SUMQ
WR ITE (6,120) AOT
WR ITE (6,140) EXITFR
                                                                                                                                                                                                                                                                                                                                          AWTST(I)=-WTST(I)
WRITE (6,150) I,AWTST(I)
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SIDO, 0001
SIDO, 0001
SDO
SDO
XWZ
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DTCND2=0.
UPCOND=UBARW
GO TO 40
                                                                                                                                                                                                                                                                                                                                                                                                     WRITE (6,170) TNO, NORDWS
WRITE (6,180) VOID ID, BNDIAM, BL
IF (MDIAM, EQ.2) WRITE (6,210) SI
IF (MDIAM, EQ.2) WRITE (6,220) SI
IF (MPITCH, EQ.2) WRITE (6,220) XWI
IF (XWZ.6E.0.) WRITE (6,220) XWI
WRITE (6,300) VCLIC, VOL2, BUNWT
WRITE (6,320) AVTBI
WRITE (6,320) AVTBI
WRITE (6,320) HDLOSS, PHPCON
WRITE (6,320) UPCOND
WRITE (6,340) UPCOND
                                                                                                                                                                                                                                                                                                                          DO 50 I=1 I SEC
IF (WTST(I).LT.0.)
IF (WTST(I).LT.0.)
CONTINUE
   VTBI.EQ.AVTB2)
VTBI.EQ.AVTB2)
VTBI.EQ.AVTB2)
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SUMMARY OUTPUT FOR COOLER SECTION

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AO TOT = AOTC + AOT

TN OT= TNC+TNGC

UB ARGA = (UBARW* TNO+ UBARWC*TNOC! / TNO T

AD TOA = (SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC+SUMQC-SUMQC-SUMQC-SUMQC-SUMQC-SUMQC-SUMQC-SUMQC-SUMQC-SUMQC-SUMQC-SUMQC-SUMQC-SUMQC-SUMQC-SUM
                                                                                                                                                 DT COL 2 = (AVTB2C - AVT B IC ) /ALOG( (STSAT X-AVT BIC) /(STSAT X-AVTB2C))
UP COOL = SUMQC/ (AOTC * DT COL2)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                 BARWC,ADTCLR,DELPVC,TDROPC,VELC(1),VELC(IVNOC)
UMQC
SC(1)
AOTC
                                                   DTCOL2=0.
UPCOOL=UBARW(
GO TO 20
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VCLC COOLWT
AVTB IC
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EQ. AVTB 2C)
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FERENTIAL PITCH
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LUG
                                                                                                                        (103H) ** NEITHER FORCE CONSTANTS NCR MULECULAR VOLUMES ARGUMENTS. CALCULATIONS CANNOT PROCEED. **)
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ON ON THE
EACH REGRESSION
ALL
CI D=0.1685/TKDE**1.2063+0.5328/TKDE**0.1579
RT SM=(1./AM1+1./AM2)**0.5
B= (10.7-2.46*RTSM)*1.6-4
DG=B*TK**1.5*RTSM/PATM/R12**2/CID
IF (V1.eq.0.0) GO TO 40
CONTINUE
DG G=0.0043*TK**1.5*RTSM/PATM/(V1**.333+V2**.333)**2
GO TO 40
CONTINUE
WR ITE (6.50)
CONTINUE
RE TURN
FORMAT (103H0 ** NEITHER FORCE CONSTANTS NCR MULECUL/
IIST AS ARGUMENTS. CALCULATIONS CANNOT PROCEED. **)
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ISED FOR
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(I)-XIBAR)*(YN(I)-YBAR
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IF (N° LE° 0°) 6D 70 30

XI N(N) = ALOG(X)

YN (N) = ALOG(X)

YN (N) = ALOG(X)

YN (N) = ALOG(X)

IF (N° LT° 2) 6D TO 40

SX 1N=0°0

SX 1N=0°0

IF (M° LT° 2) 6D TO 40

SX 1N=0°0

IF (M° LT° 2) 6D TO 40

SX 1N=0°0

SX 1N=SX 1N+X1N( 1)

SX 1Y=0°0

SX 1Y=SX 1Y+(X1N( 1)-X1BAR)**

CONTINUE

SX 1Y=SX 1Y+(X1N( 1)-X1BAR)**

CONTINUE

B1 = SX 1Y+(X1N( 1)-X1BAR)**

CONTINUE

B1 = SX 1Y+(X1N( 1)-X1BAR)**

TN=ALOG(T)
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FUNCTION BMUFN (C,T)
VISCOSITY OF SALINE SOLUTION. RANGE OF DATA WAS 0 - 24 PERCENT
CONCENTRATION AND 40 - 210 DEGREES FARENHEIT.
R= T+455.69
BM UFN = DEXP(-0.11591155D+2*C+0.12602329D-1*C*R+0.38637378D+4*C/R+0.14606532D-2*R+0.47595941D+4/R-0.1059252566D+2)
RETURN
END
                                      PER C ENT
                                      01
                                      CHANGE OF VARIABLE IS LIMITED TO
                                               IF (ABS(XTR-X)/X.LT.0.1) GO TO 60
XTR=X*(1.+0.1*ABS(Y-T)/(Y-T)*SW)
RETURN
END
XTR=E xP((TN-A)/B1)
60 T0 50
A=1
40 XTR=X*(1.+(Y-T)*5.*SW)
C*****
CHANGE OF V'
50
                                                                                                                                                                                                                                    EUNCTION CPAFN (T, JGAS, IMIX=0 GO TO (30,20,10), JGAS
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CON 4 2250

CON 4 2250

CON 4 2290

CON 4 2310

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CON 4 2420

CON 4 2520

CON 4 
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           (T)
0.5703*T+.00012819*T**2-.000008824*T**3
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       2892E-6-.7693E-10*T1*T1*T
                                                                                                                                                                                                                           393E-5*T**2-0.3367E-9*T
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          -3+0.10282155E-5*T1*T
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   3RINE
+9651*T1-(•9
+(2•*(•00767
[C**3]*T]
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           TEAM
                                                                                                       BTU/LB-MOL
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             S
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           HEAT CAP OF ...
10 IMIX=1

20 CUNTINUE

C INERT GAS IS CO2

C** HEAT CAPACITY FOR MW = 40.1

GAS2=0.209*40.1

CP AFN=GAS2

IF (I MIX. EQ.0) RETURN

0 CO VINUE

AIR CP BTU / LB MOLE-DEGREE RANK

GAS1=7.139-0.9884E-3*T+0.1393

CP AFN=GAS1

IF (I MIX. EQ.0) RETURN

IN GAS3= (GAS1 + GAS2 ) 72.0

EN D
                                                                                                                                                                                                                                                                                                          C 02
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   HEAT FOR BR
(0.00010404979*(C**2)1+(
12610354)*(C
                                                                                                                                                                                                                                                                                                                                                                                                                                                              20,20,30
IFIC HEAT FOR
(-0.24618473E-
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          3-6.
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IS FROM
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                                                                                                                                                                                                                                                                                                                                                                                                                                                 ** FUNCTION CPFN (
IF (C-0.05) 20
CP=1.0121559+(-1.0.00)
** EQUATION SPECIF
CP=1.0121559+(-1.0.00)
** EQUATION SPECIF
CP=.96946859+(2.0.00)
L591*C*11.55
2*(C**3))+(2.*(-1.5.00)
END
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       UNCTION HEGEN
FGEN=1093.88-0
FTURN
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TURN
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EQUATIONS
R-R-132
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END SUBROUTINE PRS DRP (TSAT, VMIX, WS, WNC, AXC, SDO, SF, DELPTP, ENHF) C PRS DRP REBUILT ON 9-16-69 TO USE EQUATION FOR SF GS TAR= (WS+WNC) / (AXO*3600.) AN RE= (SO3*66STAR)/SMUFN(TSAT) SF = (0.102+52.2 / ANRE)*ENHF DELPTP=SF*GSTAR**2*VMIX/(72.0*SG) RETURN END	FUNCTION PSATFN (T) PSATFN=2.718**(14.150119-(6452.5621/T)-(837533.21/T**2)) RETURN END	FUNCTION ROEFN (C, T) DENSITY OF SALINE SOLUTION. RANGE OF DATA WAS 0 - 26 PERCENT CONCENTRATION AND 40 - 300 DEGREES FARENHEIT. ROEFN=0.62707172E2+0.49364088E2*C-(0.4355304E-2+0.32554667E-1*C+(10.46076521E-4-0.63240299E-4*C)*T)*T RETURN END	FUNCTION SKBFN (C,T) THERMAL CONDUCTIVITY OF SALINE SCLUTION. RANGE OF DATA 0 - 24 PERCENT CONCENTRATION AND 40 - 300 DEGREES FARENHEIT SKBFN= (.30157913+.697989E-3*T12506E-5*T**22072E-10*T**31*(161709*C+1.) RETURN END	FUNCTION SMUFN (T) SMUFN=1.0E-5*(0.122+(1.001E-3)*T+(2.892E-7)*T**2-(7.693E-11)*T**3) RETURN END	FUNCTION TSATEN (P) AA=(ALGG(P)-14.150119)
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FUNCTION VGFN (T,P)
X=ALOG(I/P)
VGFN=E XP(((.103758E-2*X-.0177861)*X+1.10267)*X-.72240)
RE TURN
END COMPLEX ROOTS (1H1,37HSUBROUTINE TSATFN FINDS DE T=6452.5621**2-(4.0*AA*837533.21)
IF (DET) 10,20,20
WRITE (6,60)

CALL EXIT

X= (-6452.5621+SQRT(DET))/(2.0*AA)

Y= (-6452.5621-SQRT(DET))/(2.0*AA)

IF (X-Y) 30,40,40

ISATEN=Y

GO TO 50

ISATEN=X

CONTINUE

RETURN

FORMAT (1H1,37HSUBROUTINE TSATEN FINDS SUBROUTINE SWITCH (A,N)
DI MENSIGN A(N)
NN = N / 2
K = N + 1
DO 10 1=1,NN
T = A(I)
A(I) = A(K-I)
CONTINUE
RETURN
END

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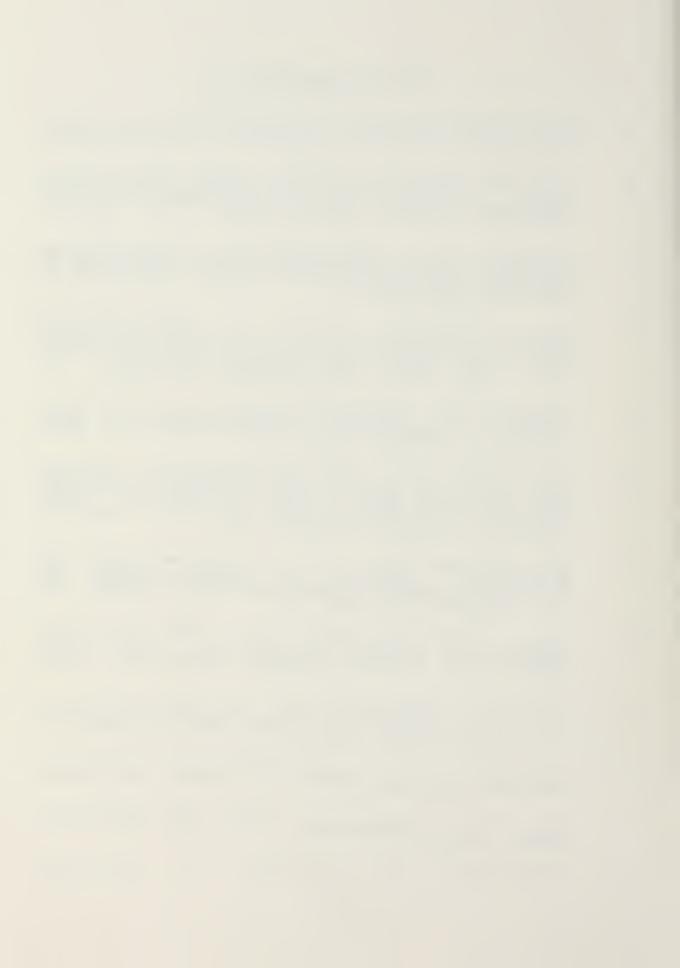
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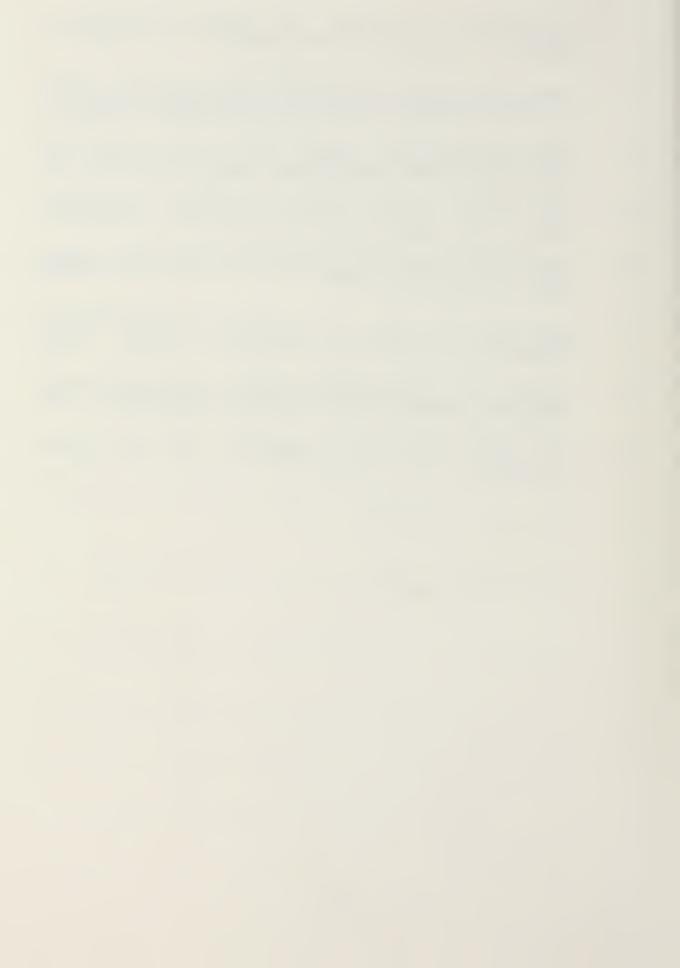
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