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NAVAL SURFACE WEAPONS CENTER WHITE OAK LAB SILVER SP--ETC F/G 17/1
SENSOR POSITION LOCATOR.(U)

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NSWC TECHNICAL REPORT

WHITE OAK LABORATORY

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SENSOR POSITION LOCATOR

BY
Jonathan Valvano

September 1976

NAVAL SURFACE WEAPONS CENTER ✓
WHITE OAK LABORATORY
SILVER SPRING, MARYLAND 20910

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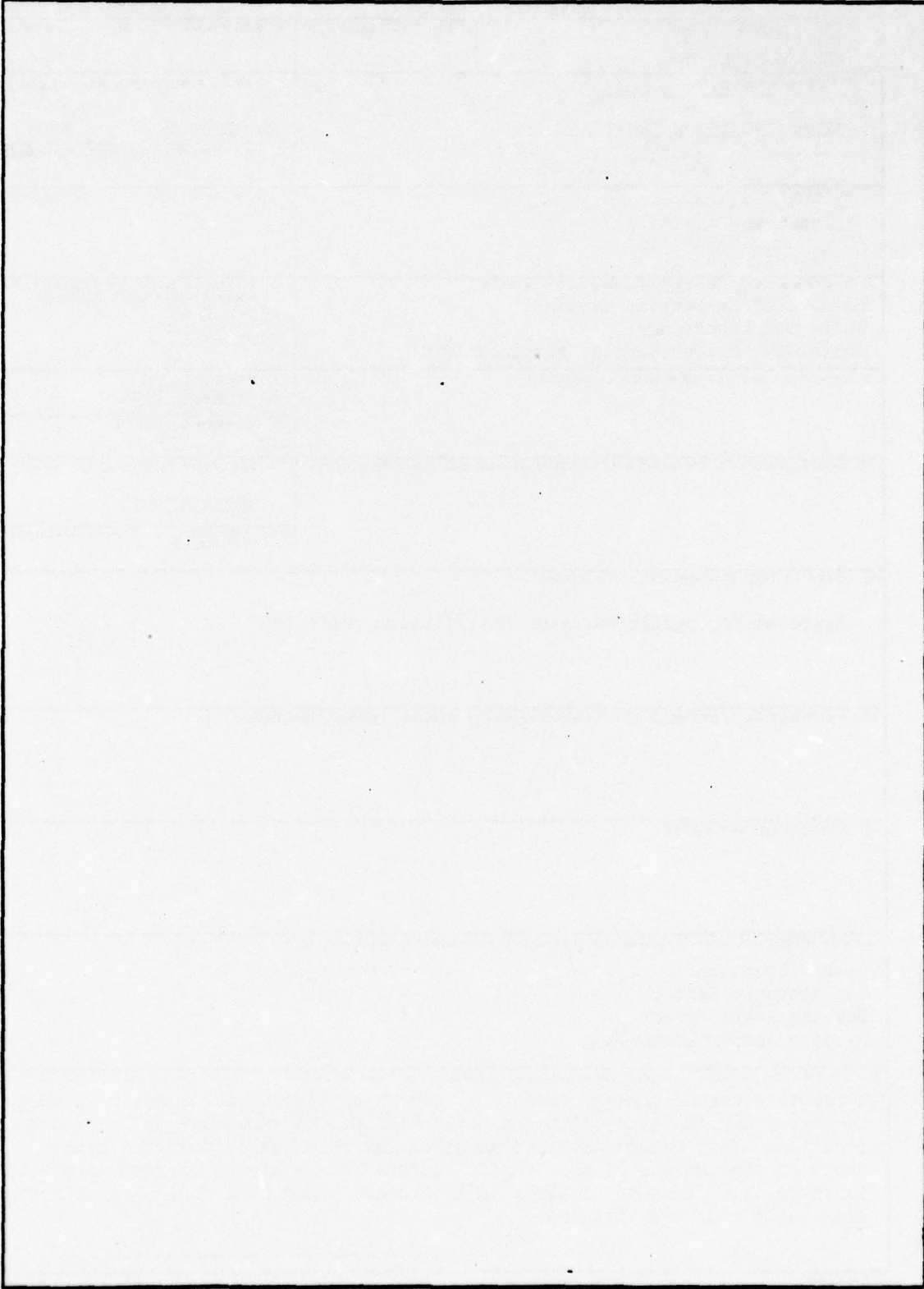
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PREFACE

This report documents a portion of the work performed at the Naval Surface Weapons Center, White Oak, Silver Spring, Maryland on the Multiple Target Classification and Location task for the REMBASS Advanced Development under Agreement Number 75-14.

C. A. Fisher
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Chapter 1

INTRODUCTION

Remote unattended ground sensors are being developed which have the capability of passively determining the bearing angle of the targets they detect and reporting this information in real time to a central location via an RF link. For one reason or another, it is not always known just where the reporting sensors are located in an exact geographical sense. Although each sensor possesses a unique identifier in the message code transmitted, the original emplacement of a sensor field may not be done by surveying them in, but rather by dropping them out of aircraft, or using rocket or gun delivery.

The purpose of this task has been to determine the location of such sensors after they have been emplaced. The proposed method is to move a target, whose location as a function of time is known, over the sensor field. The sensors will each report the bearing angle of the target several times as the target passes by. Since the target is moving, each report by a sensor will (usually) be somewhat different. That is, the sensor will report a series of angular measurements, which will differ as the target changes location. By knowing the target location as a function of time back at the sensor reporting unit the present algorithm will use the series of reports to calculate the sensor position. The method used constructs radials from the target's position towards the sensor. Each of the many radials should intersect at the sensor position, as shown in Figure 1.

A weighted least-squares algorithm is used to fit the lines to the point which represents the sensor position. This theory is relatively simple, but there are problems which complicate the calculations.

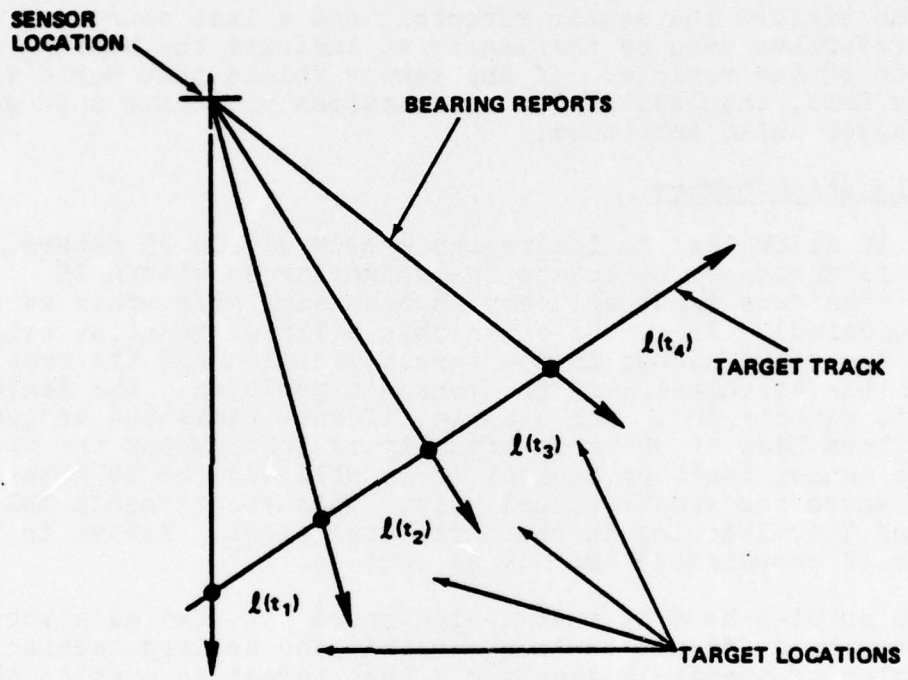


FIG. 1 SENSOR FIELD

Chapter 2

PROBLEMS AND SOLUTIONS

As stated there are several factors which complicate the calculation of a sensor's position. One of these is the uncertainty of the location and speed of the "known" target used to locate the sensor. A second problem occurs due to extraneous noise sources which can distort the sensor reports. And a last source of error is the reference used by the sensor to indicate the bearing or direction of the vehicle. If the sensor thinks that North is actually East, then all of its calculations will have a 90 degree error factor built into them.

Target Location Errors

If it is desired to locate the sensor within 25 meters, then it is necessary to locate the known target within 25 meters (even more input accuracy is necessary when other errors are considered). It can be shown that a linear relation exists between the planar error in the target location and the resulting error in the calculation of the sensor's position. The scale factor is exactly one. For example, if one thinks the target is 20 meters East of where it actually is (throughout the track) then the Sensor Position Locator (SPL) will also be 20 meters East of where the sensor actually is. This relationship holds for X and Y positioning in the horizontal plane. Errors in the altitude (Z coordinate) are not as serious.

The problem here is that a high speed jet used as a known target may be difficult to locate within the desired accuracy. The problem of precisely locating a test target is outside the scope of this investigation, but the errors it causes are not. It is evident that the available accuracy for sensor position location is determined by the current capabilities of target position location for cooperating targets.

Sensor Report Errors

At the time of investigation, the sensors themselves are also under development. It is not known exactly what kind of accuracy these sensors will have, and the bearing values reported

by the sensors have to be believed, to some extent, as the truth. Therefore, errors originating in the sensor will propagate to the SPL algorithm. It is not the purpose of this task to reduce these inherent sensor errors, but rather to reduce the propagation of these errors to the greatest possible extent. Associated with these "sensor errors" are errors caused by other noise sources in the environment.

It is assumed that the magnitude of the error does fluctuate and the simulation in this work has used gaussian noise as an error model. It is reasonable to try to catch and use the data when it is close to error-free and ignore the data when it is believed to contain large errors. The problem is therefore to find some indicator to tell us of the magnitude of the error, and such indicators do exist. Possibilities include signal to noise ratio, signal strength, and distance from the sensor to the target. The first and last of these indicators were used in the simulation and favorable results were obtained.

The least-squares algorithm will accept a weighting factor (W) based on data quality and it appears that full use of it will improve the quality of the SPL.

Because of the multi-pass SPL algorithm, it is possible to estimate the radial distance from the sensor to the target, and then use it to construct a weighting factor within the algorithm itself. This too has been done in the simulation and improvements were noted.

Sensor Rotation Error

In addition to assuming that the sensor will fall in an arbitrary (X, Y) position, it is also assumed that the sensor will have an arbitrary angular orientation. When the sensor reports, "I see a target at 90 degrees," does that mean North, East, South, or West?

It is probable that each sensor will ultimately contain its own compass, in which case, this source of error would be minimized. It seems reasonable that errors in angular orientation calculated inside the sensor might be on the order of five degrees, when one considers that the compass will be in close proximity to metal parts of the sensor, etc. This amount of angular error, called angle bias, can produce very unsatisfactory errors in the SPL

simulations (on the order of hundreds of meters). Therefore, it seems necessary to include a bias calculation within the SPL algorithm. Fortunately, it is possible to calculate the angle bias without knowing the sensor position, and the latter can then be determined using this bias for compensation.

Errors in the SPL simulation due to all of these sources are discussed in more detail in Appendix A.

Chapter 3

MATH MODEL FOR CALCULATION OF ROTATION BIAS

This is a straightforward algorithm that uses three bearing reports ($\alpha_1, \alpha_2, \alpha_3$) that all have the same bias. It is not necessary that the reports be sequential, and in fact if there are many reports which differ only slightly in bearing, then it is advisable not to use sequential reports because they will probably be in the same or adjacent bearing bin. The algorithm uses angle differences in which the bias subtracts out and, using the geometry of adjacent triangles, calculates an internal angle which in turn leads to the bias. The algorithm will not work if the angle differences are zero which occurs if the target has not moved.

Figure 2 shows the geometry necessary for the calculation of the rotational bias of the sensor. If the sensor reports the three angular bearings ($\alpha_1, \alpha_2, \alpha_3$) shown in Figure 2, then d_{12} and d_{23} are the linear distances along the target track between those reports. α_A is the direction of the target track with respect to North. By applying the law of sines to Figure 2 one obtains equations 1 and 2 and by also applying Figure 2, the law that the sum of the interior angles of a triangle is 180 degrees, one obtains equations 3 and 4.

$$\frac{d_{12}}{\sin \theta_1} = \frac{d_{S2}}{\sin A} \quad (1)$$

$$\frac{d_{23}}{\sin \theta_2} = \frac{d_{S2}}{\sin D} \quad (2)$$

$$\theta_1 + \theta_2 + A + D = \quad (3)$$

$$\alpha_A + (\pi - A) + (\pi - \text{Bias} - \alpha_1) = \pi \quad (4)$$

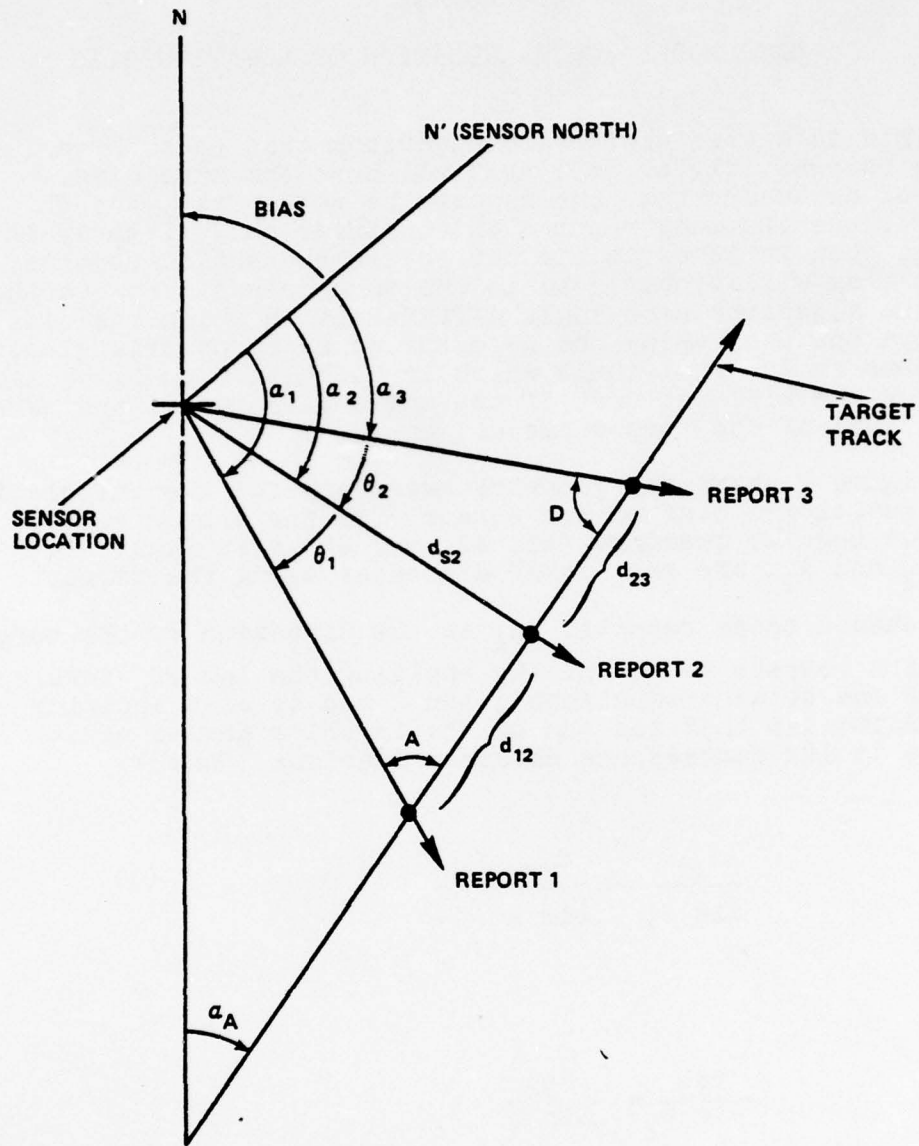


FIG. 2 BIAS CALCULATION - TRIANGLES

Eliminating d_{S2} from equations 1 and 2 yields:

$$d_{12} \frac{\sin A}{\sin \theta_1} = d_{23} \frac{\sin D}{\sin \theta_2} \quad (5)$$

Using equations 3 and 5 to eliminate D and then solving for A one finds:

$$A = \arctan \left[\frac{\sin (\theta_1 + \theta_2)}{\frac{d_{12}}{d_{23}} \cdot \frac{\sin \theta_2}{(\sin \theta_1)} - \cos (\theta_1 + \theta_2)} \right] \quad (6)$$

$$\text{where } \theta_1 = (\alpha_1 - \alpha_2)$$

$$\theta_2 = (\alpha_2 - \alpha_3)$$

Equation 4 can be rewritten to solve for the rotational bias:

$$\text{BIAS} = \pi + \alpha_A - \alpha_1 - A \quad (7)$$

Referring to Figure 2, it is possible for the rotational bias to be in the other direction. That is, for the line N' to be on the other side of N. If this is the case, the rotational bias from equation 7 will turn out to be negative.

Least Squares Method for Bias Calculation

Due to the inherent errors in any system, the results achieved by this calculation of the rotational bias will be in error by some amount. If one was to perform this calculation several times for different target runs one would get several values for the rotational bias, called B_1 . Assuming the errors are gaussian, a best value for the rotational bias would be the mean and an indication of the accuracy of that value would be the standard deviation. If some values are known to be more accurate than others, they can be multiplied by some weighting factor, W_1 , to yield better results. Then the final value for the bias would be:

$$\text{BIAS} = \frac{\sum_{i=1}^n (W_i B_i)}{\sum_{i=1}^n (W_i)} \quad (8)$$

The standard deviation for this BIAS calculation is:

$$\sigma = \frac{\sum_{i=1}^n W_i (B_i^2 - \text{BIAS}^2)^{1/2}}{\sum_{i=1}^n W_i} \quad (9)$$

Effects of the Speed of Sound

It is easily observed, when listening to an approaching high-speed target, that the target is ahead of the incoming sound waves. This is because the sound takes a finite Δt seconds to travel from the target to the observer. In this Δt seconds the target moves, and is therefore ahead of where it was when the sound originated.

The point is to determine the effect of this error on the calculation of the rotational bias and to correct for it. In two dimensions this error can be deterministically calculated independent of radial distance and therefore can be calculated without knowing the sensor's location.

Correction Algorithm. Let Δt be the time elapsed between the time when the sound originated at the target (t_1) and the time when the sound is received at the sensor (t_2). Figure 3 depicts this case where C is the speed of sound, V is the velocity of the target, $C(\Delta t)$ is the distance from the sensor to the target at time t_1 , $V(\Delta t)$ is the distance the target will travel in the time Δt , α_A is the target course, θ is the angular error due to the speed of sound, and α_{t1} is the report vector referenced to true North (not sensor North). Applying the law of signs (equation 12) and the sum of the interior angles of a triangle equals 180 degrees (equations 10 and 11) to Figure 3 yields:

$$(180 - \alpha_{t1}) + (180 - \beta) + \alpha_A = 180^\circ \quad (10)$$

$$\gamma = 180^\circ - \theta - \beta \quad (11)$$

$$\frac{C(\Delta t)}{\sin \gamma} = \frac{V(\Delta t)}{\sin \theta} \quad (12)$$

Substituting from equation 11 into equation 12 and simplifying gives the error due to the speed of sound, θ , to be:

$$\theta = \arctan \frac{\sin \beta}{\frac{C}{V} - \cos \beta} \quad (13)$$

And from equation (10) we find β is:

$$\begin{aligned} \beta &= 180 - \alpha_{t1} = \alpha_A \\ &= 180 \text{ degrees} - \text{report bearing referenced to true} \\ &\quad \text{North} + \text{target course.} \end{aligned}$$

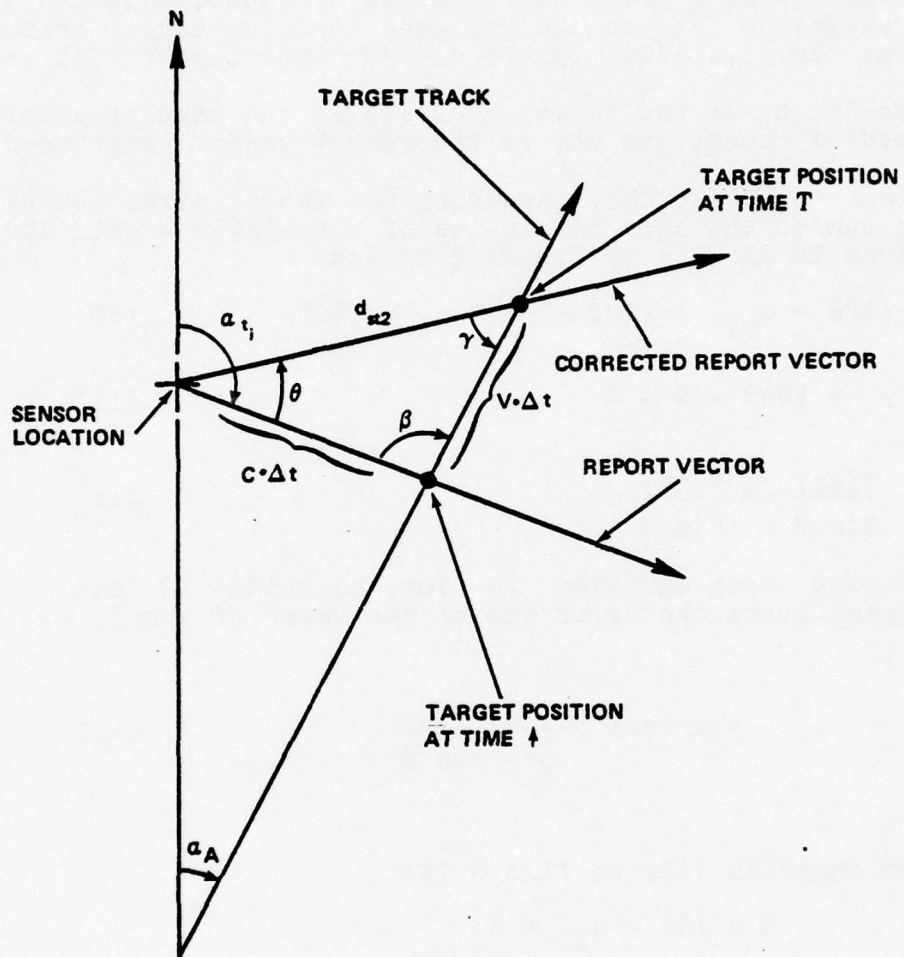


FIG. 3 SPEED OF SOUND CORRECTION IN 2-D

Figure 4 gives the correction angles for some values of B and V/C. As an example, if the target were reported such that B = 60 degrees and the velocity was 300 mph, or V/C = 0.4; then the target would really be 23 degrees ahead of the report bearing.

Adjustments for 3-D. The previous discussions were all done in a two dimensional space, that is for the target track in the same plane as the sensor. To extend the algorithm to three dimensions will require the following simplifying assumptions; (1) that the target will travel in a horizontal plane parallel to the ground plane, (2) that the sensor will report in the ground plane as if the target were also in the ground plane, and (3) that the altitude of the sensor, Z₀, will be assumed to be zero. Thus the altitude of the target, Z_A, is just the perpendicular distance between the two parallel planes. Figure 5 depicts these assumptions.

Correction Due to Speed of Sound in 3-D. In 2-D the correction error, θ , due to the speed of sound is independent of the horizontal distance from the sensor to the target track. Unfortunately, this is not true if there is a sizable altitude difference between the target track and the sensor.

Since θ is needed to calculate the sensor location and the sensor location is needed to calculate the altitude correction to θ , an iterative algorithm must be used whereby the sensor location is first calculated without the altitude correction for θ . This sensor location is then used to determine a more accurate value for θ , this new value of θ is used to generate an improved sensor location, and so forth until the desired accuracy is achieved.

Figure 6 shows the speed of sound correction for three dimensions where Z_A is the perpendicular distance between the target plane and the sensor plane, RXYT₁ is the distance from the sensor to the projection of the target onto the sensor plane at time t₁, and θ is the correction error due to the speed of sound. Note that both γ and θ are in the ground plane.

The calculations for the three dimensional case are very similar to those for the two dimensional case. In fact, equations 10 and 11 still hold. However, equation 12 now reads:

$$\frac{RXYT_1}{\sin \gamma} = \frac{(V) (\Delta t)}{\sin \theta} \quad (15)$$

V/C OR RATIO	.9	.8	.7	.6	.5	.4	.3	.2	.1	.05	.025	.0125	0	
ANGLE β	MPH	675	600	525	450	375	300	225	150	75	37.5	18.8	9.4	0
	M/S	302	268	235	201	168	134	101	67	34	17	8.4	4.2	0
180°		0°	0°	0°	0°	0°	0°	0°	0°	0°	0°	0°	0°	0°
150°		14°	13°	12°	11°	10°	8°	7°	5°	3°	1°	1°	0°	0°
135°		21°	20°	18°	17°	15°	12°	10°	7°	4°	2°	1°	1°	0°
120°		32°	26°	24°	22°	19°	16°	13°	9°	5°	2°	1°	1°	0°
90°		42°	39°	35°	31°	27°	22°	17°	11°	6°	3°	1°	1°	0°
60°		55°	49°	43°	37°	30°	23°	17°	11°	5°	3°	1°	1°	0°
45°		60°	52°	44°	36°	27°	22°	15°	9°	4°	2°	1°	1°	0°
30°		64°	52°	42°	32°	24°	17°	11°	7°	3°	1°	1°	0°	0°
10°		54°	33°	21°	14°	10°	7°	4°	2°	1°	1°	0°	0°	0°
0°		0°	0°	0°	0°	0°	0°	0°	0°	0°	0°	0°	0°	0°

FIG. 4 SAMPLE BEARING ERROR DUE TO THE SPEED OF SOUND

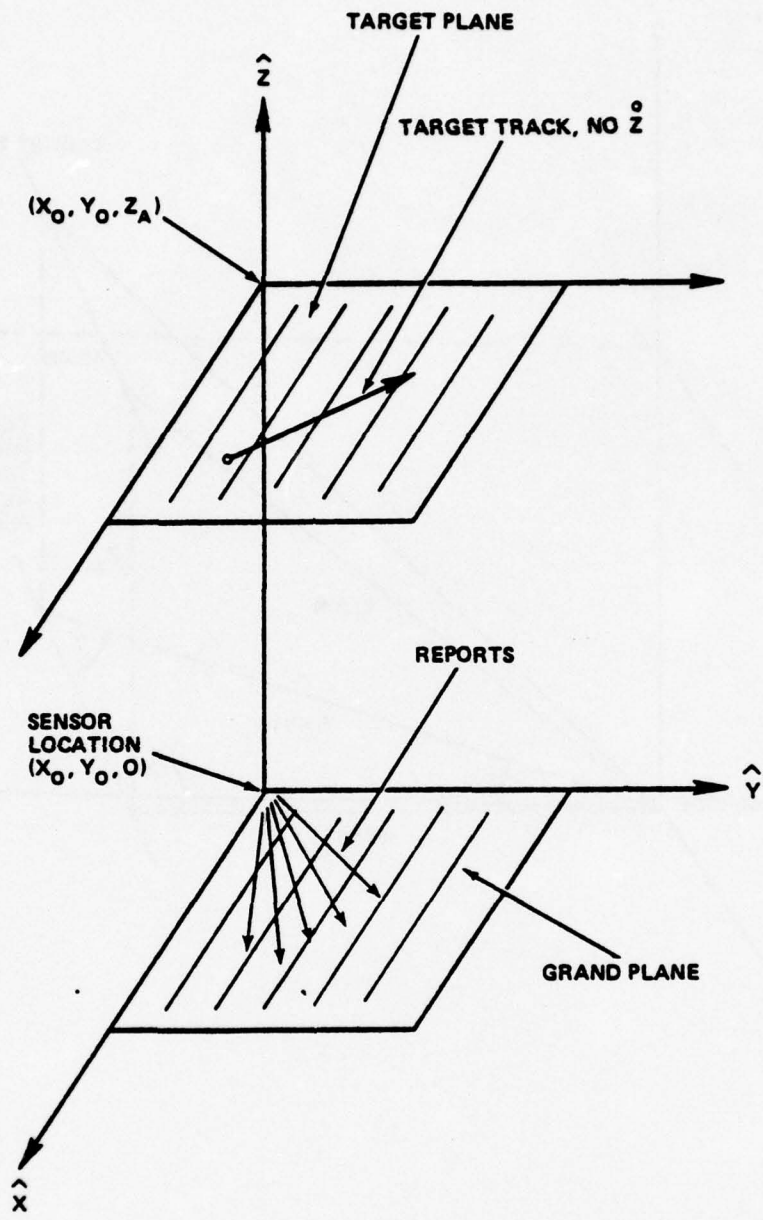


FIG. 5 ASSUMPTIONS IN THE 3-D MODEL

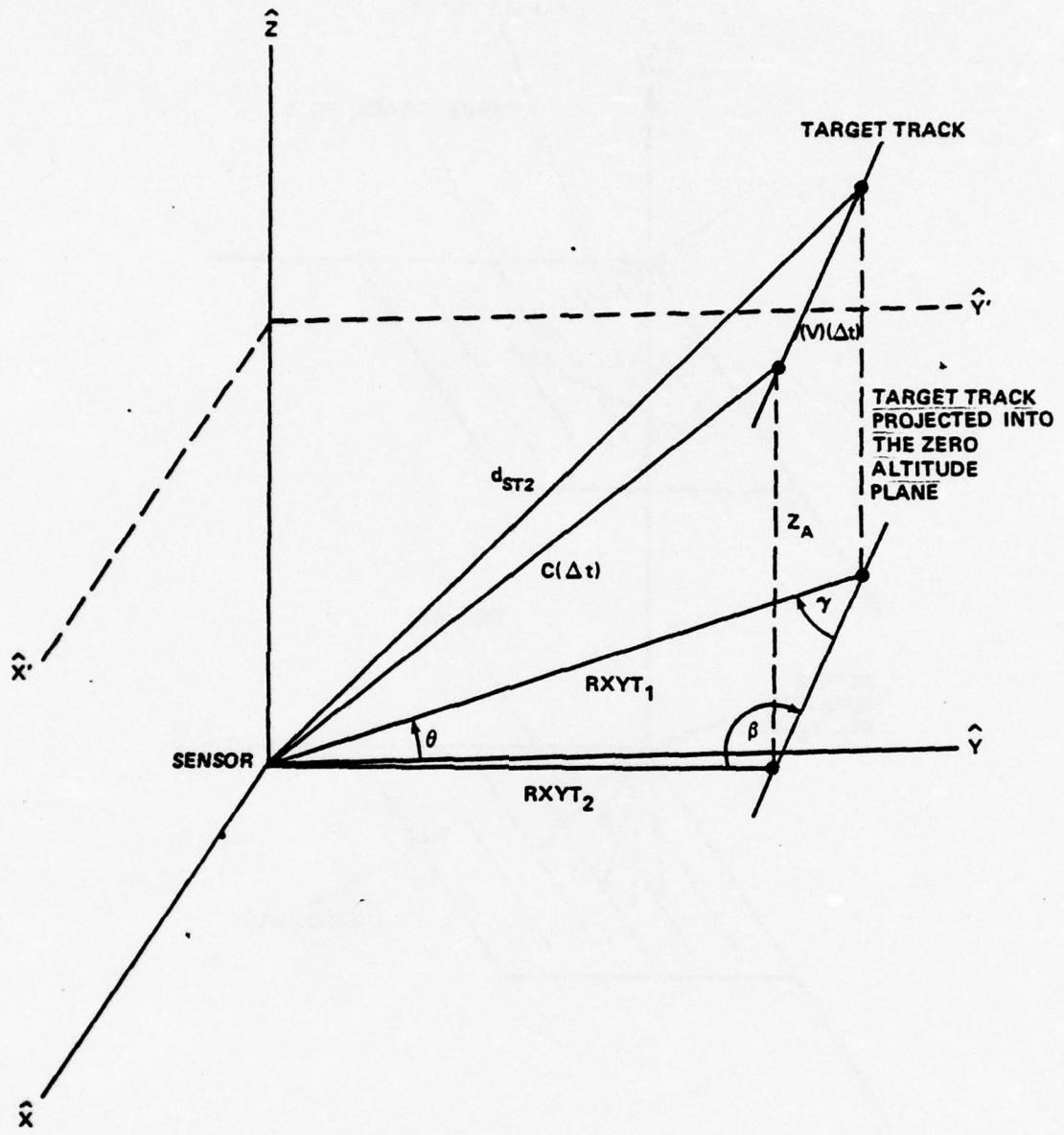


FIG. 6 SPEED OF SOUND CORRECTION IN 3-D

Observe that the projection of the line (C) (Δt) onto the sensor plane forms a right triangle. This gives one a relationship between equations 12 and 15, namely:

$$(C) (\Delta t) = \sqrt{(ZA)^2 + (RXTT_1)^2} \quad (16)$$

$$\text{Let Ratio} = \left(\frac{C}{V}\right) \left(\frac{1}{\sqrt{1 - (ZA/RXYT_1)^2}}\right) \quad (17)$$

Then equation 13 for the three dimensional case becomes:

$$\theta = \arctan \frac{\text{Sin } \beta}{\text{Ratio} - \text{cos } \beta} \quad (18)$$

Unfortunately, to calculate θ , one needs to know $RXYT_1$ (equation 17), and $RXYT_1$ is dependent upon the sensor location.

Calculating $RXYT_1$. Chapter 4 discusses the sensor position. Once this is done one can come back and calculate $RXYT_1$.

Figure 7 is a two dimensional view of the ground plane of the three dimensional speed of sound correction shown in Figure 6. β is defined by equation 14, θ is the correction error due to the speed of sound, and $RXYT_1$ and $RXYT_2$ are the projections of (C)(Δt) and (d_{st2}) onto the sensor plane.

From the definition of the sine of a right triangle it is seen from Figure 7 that:

$$RXYT_1 = \frac{\sqrt{(X_0 - X_p)^2 + (Y_0 - Y_p)^2}}{\text{Sin } \beta} \quad (19)$$

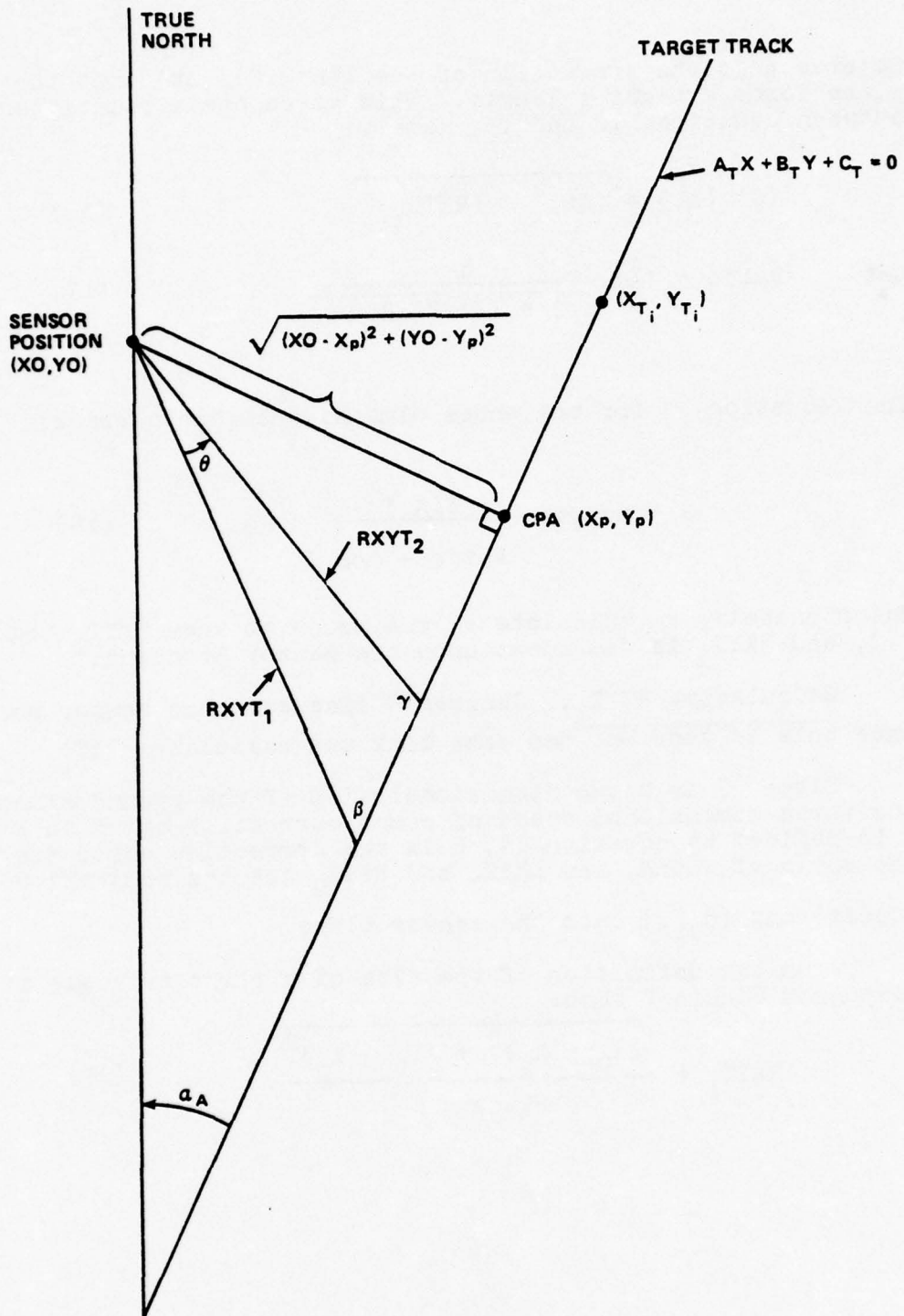


FIG. 7 CALCULATION OF R_{XYT_1}

The value $\sqrt{(X_0 - Y_p)^2 + (Y_0 - Y_p)^2}$ is simply the distance between the two points (X_0, Y_0) and (X_p, Y_p) . The point (X_0, Y_0) is the sensor location which is calculated in Chapter 4. The other point (X_p, Y_p) is calculated as the point on the target track closest to the sensor.

The equation of the line ($A_T X + B_T Y + C_T = 0$) can be found from any point (X_{T_1}, Y_{T_1}) on that line and from the slope (α_A) of that line. From standard algebraic formulas:

$$A_T = \cos (\alpha_A) \quad (20)$$

$$B_T = \sin (\alpha_A) \quad (21)$$

$$C_T = A_T X_{T_1} - B_T Y_{T_1} \quad (22)$$

The co-ordinates of the point (X_p, Y_p) on the line $A_T X + B_T Y + C_T = 0$ that is closest to the point (X_0, Y_0) is given by:

$$X_p = \frac{-(A_T C_T + B_T X_0)}{(A_T^2 + B_T^2)} \quad (23)$$

$$Y_p = \frac{A_T (A_T Y_0 - B_T X_0) - B_T C_T}{(A_T^2 + B_T^2)} \quad (24)$$

Now $RXYT_1$ (equation 19) can be calculated and substituted into equations 17 and 18.

Chapter 4

CALCULATION OF SENSOR POSITION

At this point it is assumed that the reported bearing has been corrected and that it points from the sensor to the target at time t_2 . The method of approach is to calculate the parameters of a line, $L_1 = a_1x + b_1y + c_1$, passing through the target and along the reported bearing ray. This is done for every report. Therefore, many lines will be calculated, all of which should intersect at the sensor's position. Because of the inaccuracies in the bearing report and in the target's position it is not expected that every line will intersect exactly at the sensor's position. Figure 8 shows a typical case.

The idea is to find a point (X_0, Y_0) which best fits the intersection of all the lines. A weighted least-squares algorithm is used, and the distance function is the ratio of the perpendicular distance, d , from the point (X_0, Y_0) to each line over the radial distance, rx_1 , (where rx_1 is equivalent to $RXYT_2$ in Figure 6) from the target to the sensor. This ratio is the arc sine of theta, θ_1 , as shown in Figure 9 for one bearing report. For small θ ,

$$\theta_1 = \frac{d_1}{rx_1} \quad (25)$$

Math Model for SPL

The weighted least-squares algorithm to be used requires minimizing the equation below where W_1 is a weighting function:

$$S = \sum_{i=1}^n W_1(\theta_1)^2 \quad (26)$$

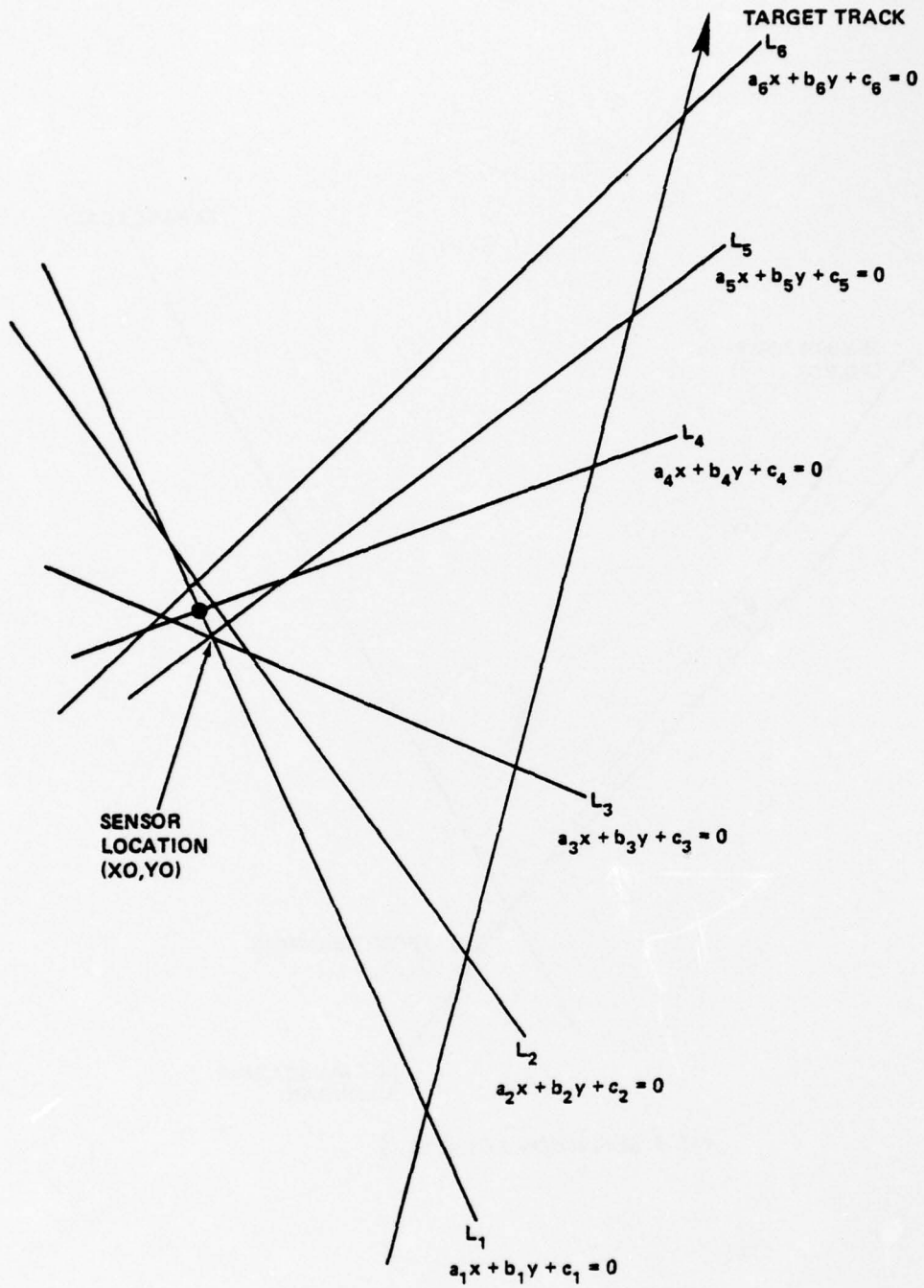


FIG. 8 SENSOR POSITION LOCATOR

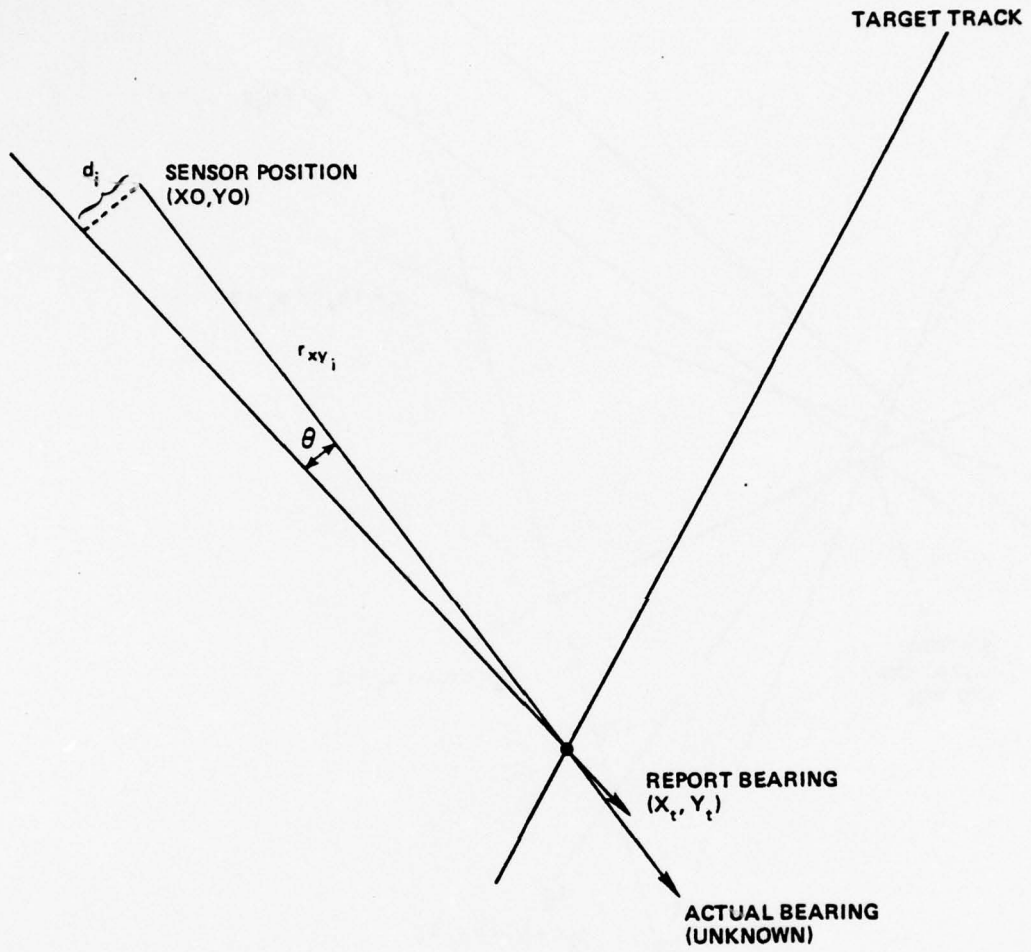


FIG. 9 DEFINITION OF THETA, θ

The distance from the sensor location, point (X_0, Y_0) , to the line L_i is given by:

$$d_i = \left| \frac{a_i X_0 - b_i Y_0 + C_i}{a_i^2 + b_i^2} \right| \quad (27)$$

The length of the line rx_{y_i} is given below where (X_0, Y_0) is the sensor location and (X_{T1}, Y_{T1}) is the i^{th} target location.

$$rx_{y_i} = \sqrt{(X_0 - X_{T1})^2 + (Y_0 - Y_{T1})^2} \quad (28)$$

Substituting equations 25 and 27 into equation 26, one gets:

$$S = \sum_{i=1}^n \frac{W_i (a_i X_0 + b_i Y_0 + C_i)^2}{(a_i^2 + b_i^2) (rx_{y_i})^2} \quad (29)$$

To minimize equation 29 one needs to take the partials with respect to X_0 and Y_0 . To simplify these calculations let X_0 and Y_0 in equation 28 be the sensor position from a previous calculation (\bar{X}_0, \bar{Y}_0) . Then rx_{y_i} is not a function of X_0 and Y_0 and can be treated as a constant. The above simplifying assumption results in acceptable errors. Without this assumption the model cannot be brought into closed form. A more complex derivation has been worked out but not implemented here. It was abandoned because of the fast growth of complexity with little promise of any accuracy to be gained.

Taking the partials of equation 29 with respect to X_0 and Y_0 and setting them equal to zero gives the following formulas for calculation of X_0 and Y_0 :

$$\begin{aligned}
 x_0 = & \left(\frac{\sum w_1 b_1 c_1}{(a_1^2 + b_1^2) RXY_1^2} \right) \left(\frac{\sum w_1 a_1 b_1}{(a_1^2 + b_1^2) RXY_1^2} \right) - \left(\frac{\sum w_1 a_1 c_1}{(a_1^2 + b_1^2) RXY_1^2} \right) \left(\frac{\sum w_1 b_1^2}{(a_1^2 + b_1^2) RXY_1^2} \right) \\
 & \left(\frac{\sum w_1 a_1^2}{(a_1^2 + b_1^2) RXY_1^2} \right) \left(\frac{\sum w_1 b_1^2}{(a_1^2 + b_1^2) RXY_1^2} \right) - \left(\frac{\sum w_1 a_1 b_1}{(a_1^2 + b_1^2) RXY_1^2} \right)^2
 \end{aligned}
 \tag{30}$$

$$\begin{aligned}
 y_0 = & \left(\frac{\sum w_1 a_1 c_1}{(a_1^2 + b_1^2) RXY_1^2} \right) \left(\frac{\sum w_1 a_1 b_1}{(a_1^2 + b_1^2) RXY_1^2} \right) - \left(\frac{\sum w_1 b_1 c_1}{(a_1^2 + b_1^2) RXY_1^2} \right) \left(\frac{\sum w_1 a_1^2}{(a_1^2 + b_1^2) RXY_1^2} \right) \\
 & \left(\frac{\sum w_1 a_1^2}{(a_1^2 + b_1^2) RXY_1^2} \right) \left(\frac{\sum w_1 b_1^2}{(a_1^2 + b_1^2) RXY_1^2} \right) - \left(\frac{\sum w_1 a_1 b_1}{(a_1^2 + b_1^2) RXY_1^2} \right)^2
 \end{aligned}
 \tag{31}$$

Calculation of A_1 , B_1 , C_1

In order to evaluate equations 30 and 31 one needs to calculate the parameters A_1 , B_1 , and C_1 that form the line $A_1X + B_1Y + C_1 = 0$. This line passes through the target position (XT_1, YT_1) and has a corrected report bearing of $d + c$ as shown in Figure 10. $d + c$ is just $at_1 - \theta$. Since the slope and one point on a line determine its equation, we find that:

$$A_1 = \cos (d + c) \quad (32)$$

$$B_1 = \sin (d + c) \quad (33)$$

$$C_1 = A_1(XT_1) - B_1(YT_1) \quad (34)$$

Substituting these values and rx_{y_1} (equation 28) into equations 30 and 31 one can solve iteratively for the sensor position (X_0, Y_0) . An iterative solution is necessary since rx_{y_1} is calculated using the values of (X_0, Y_0) obtained from the previous iteration.

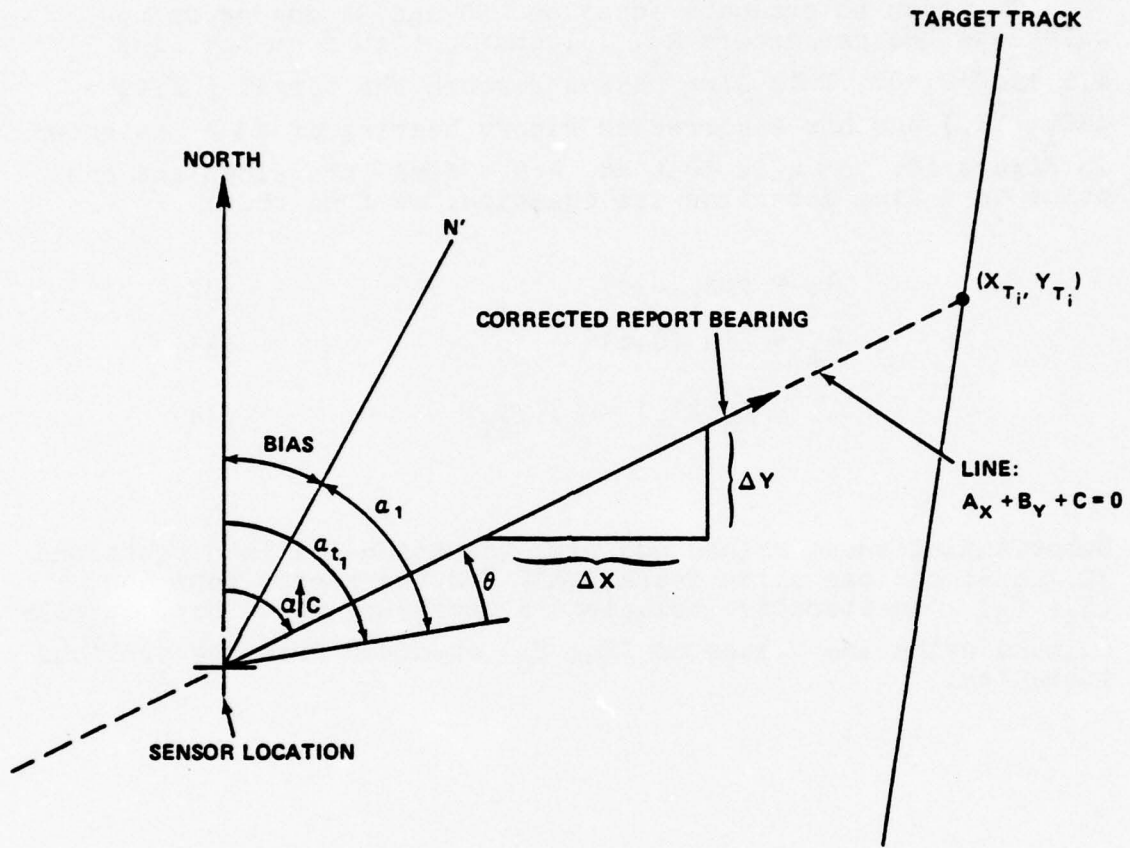


FIG. 10 CALCULATION OF A_i, B_i, C_i

Chapter 5

APPLICATION OF SPL ALGORITHM

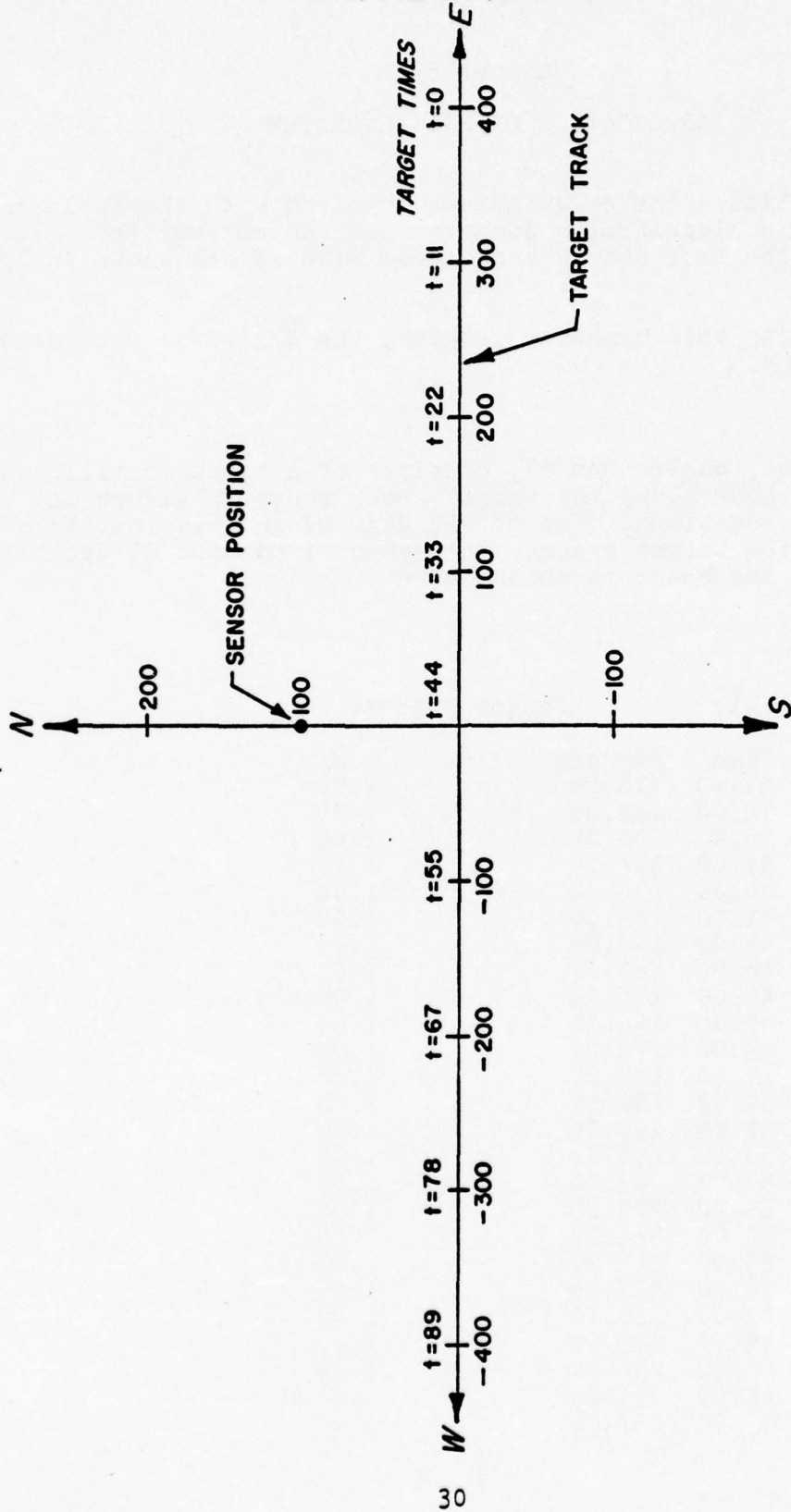
To facilitate the calculations involved with the implementation of this algorithm, a computer program was written. This program and the flow charts associated with it are shown in Appendix B.

Then using this computer program, the algorithm was tested with two trial runs.

Run #1

The first, called Run #1, consists of a truck traveling at 20 miles per hour along the target track shown in Figure 11. The sensor had a rotational bias of -25 degrees and was located 100 meters from the target track. The sensor indicated 25 detections at the times and bearings shown below.

Run #1	Sensor Reports	
Time	Bearing	W
34.00	116.00	1.00
35.00	119.00	1.00
38.00	122.00	1.00
39.00	127.00	1.00
40.00	136.00	1.00
41.00	144.00	1.00
45.00	147.00	1.00
46.00	153.00	1.00
47.00	158.00	1.00
48.00	164.00	1.00
49.00	172.00	1.00
50.00	178.00	1.00
51.00	186.00	1.00
52.00	192.00	1.00
53.00	198.00	1.00
54.00	203.00	1.00
55.00	209.00	1.00
57.00	212.00	1.00
58.00	217.00	1.00
59.00	223.00	1.00
63.00	226.00	1.00
65.00	229.00	1.00
66.00	231.00	1.00
67.00	234.00	1.00
69.00	237.00	1.00



SIMULATED RUN #1; TARGET TRACK

FIGURE II

The weighting factor used in the calculations was 1.00 for all points. When the test vehicle passed certain locations along the target track, called stakes, the time was noted. These times and the stake locations are also included in Figure 11.

The sensor position was calculated under three different conditions. For the first case, the rotational bias of the sensor was calculated and this value was used in calculating the location of the sensor. In the other two cases, a value for the rotational bias of the sensor was put into the program and this value was used to calculate the sensor location. The results of these three calculations are shown below where (X0, Y0) is the sensor location, Radial error = $\sqrt{(X0)^2 + (Y0)^2}$, and ALROT is the

sensor angular rotation bias (either calculated or externally set equal to some value.

ROTATIONAL BIAS CALCULATED

	X0	Y0	ALROT
REAL	0	100.00000	0
CALCULATED	-38.21974	86.88408	-5.29993
DIFFERENCE	38.21974	13.11592	5.29993
RADIAL ERROR	40.40762		

ROTATIONAL BIAS SET = 0 DEGREES

	X0	Y0	ALROT
REAL	0	100.00000	0
CALCULATED	-50.52054	88.81540	0
DIFFERENCE	50.52054	11.18460	0
RADIAL ERROR	51.74380		

ROTATIONAL BIAS SET -25 DEGREES

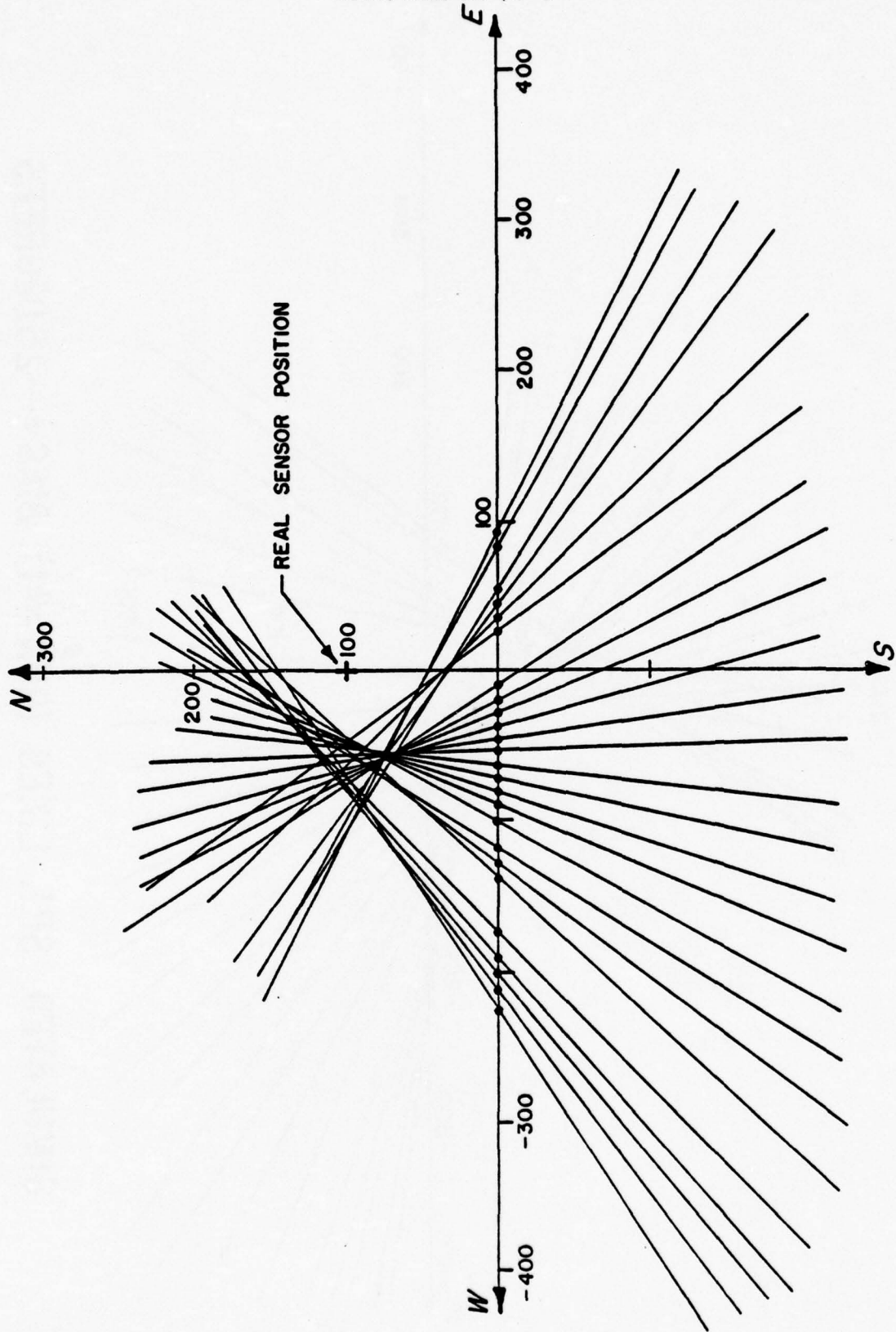
	X0	Y0	ALROT
REAL	0	100.00000	-25.00000
CALCULATED	-3.74224	103.15131	-25.00000
DIFFERENCE	3.74224	-3.15131	0
RADIAL ERROR	4.89235		

The Radial Error is lowest for the case where the rotational bias was set at -25 degrees, which indicates that the best results were achieved in this case. This is true because the actual rotational bias of the sensor was -25 degrees.

Figures 12 and 13 graphically depict the two cases where the rotational bias was externally set to either 0 or -25 degrees. In these figures lines were drawn from the vehicle location (whenever the sensor made a report) towards the sensor along a corrected bearing. This corrected bearing was just the bearing reported by the sensor plus 180° minus the rotational bias of the sensor.

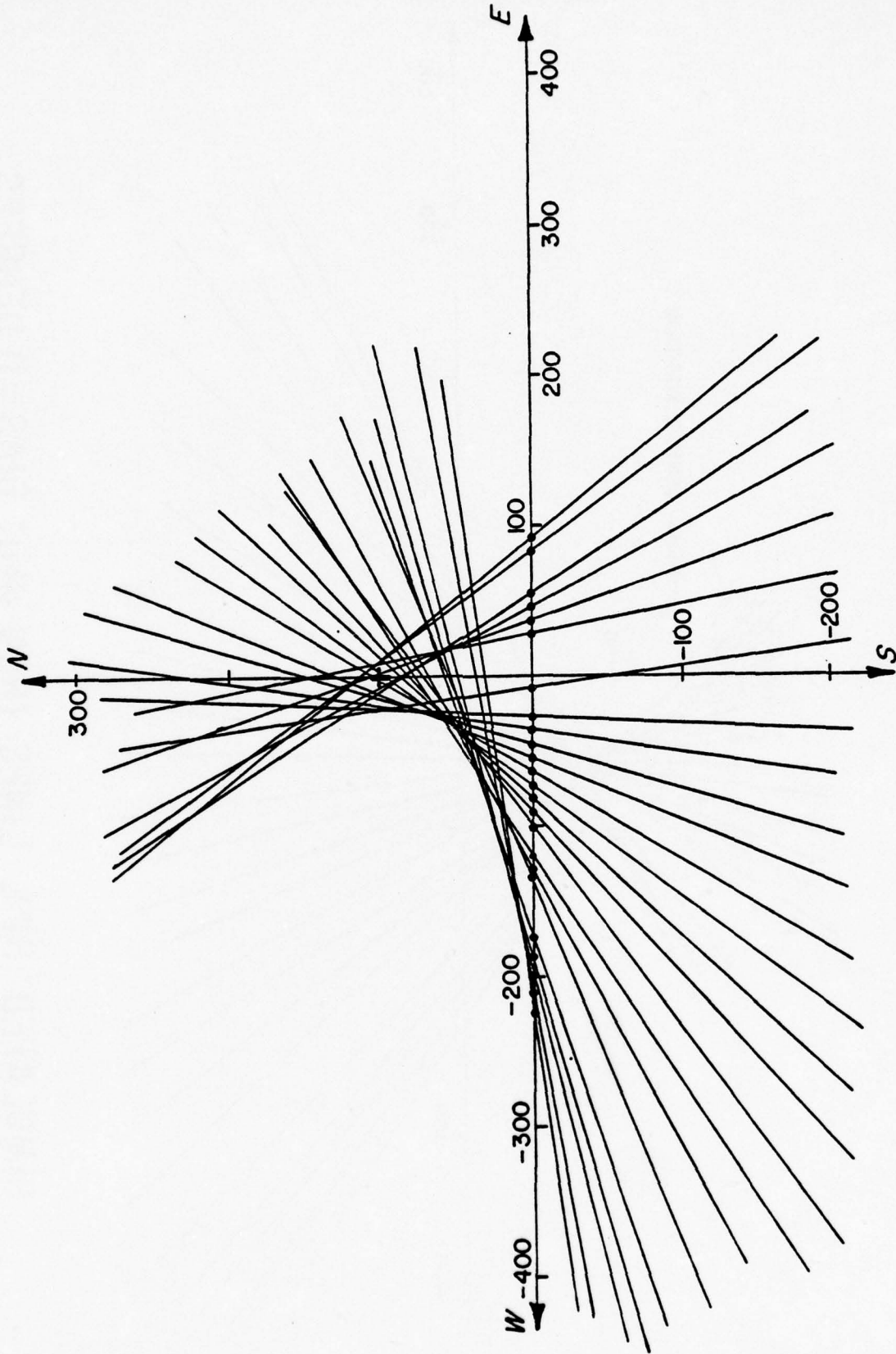
Run #2

The second run was very similar to the first. This time a truck was again driven by the sensor at a CPA of 100 meters, and with a velocity of 9 meters/second. However, this time the sensor had a rotation bias of 9 degrees, and the sensor was placed on the other side of the target track. Figure 14 shows the target track and the times that the vehicle passed the different stake locations.



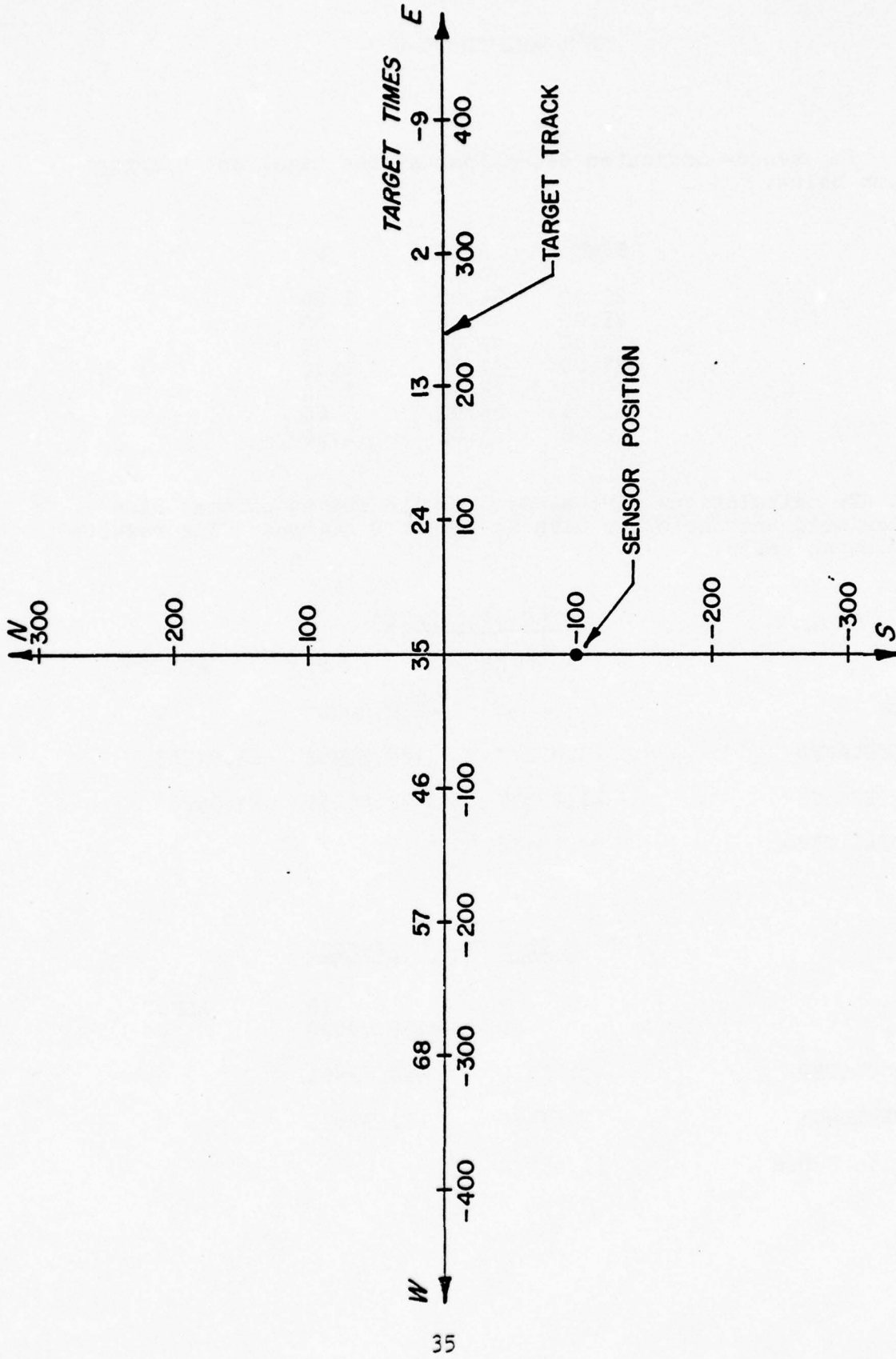
SIMULATED SPL LINES (RUN #1); BIAS = 0 DEGREES

FIGURE 12



SIMULATED SPL LINES (RUN #1); BIAS = -25 DEGREES

FIGURE 13



SIMULATED RUN #2; TARGET TRACK

FIGURE 14

The sensor indicated detections at the times and bearings shown below.

TIME	BEARING	W
23.00	51.00	1.00
25.00	46.00	1.00
28.00	43.00	1.00
29.00	37.00	1.00
30.00	34.00	1.00
31.00	26.00	1.00
32.00	23.00	1.00

Two SPL calculations were made; one with the rotational bias calculated and the other with it set at 0 degrees. The results are shown below:

BIAS CALCULATED

	X0	Y0	ALROT
REAL	0	-100.00000	0
CALCULATED	1.47177	-67.59963	-23.99265
DIFFERENCE	-1.47177	-32.40037	23.99265
RADIAL ERROR	32.43378		

BIAS SET = 0 DEGREES

	X0	Y0	ALROT
REAL	0	-100.00000	0
CALCULATED	-3.67670	-86.62461	0
DIFFERENCE	3.67670	-11.37539	0
RADIAL ERROR	11.95482		

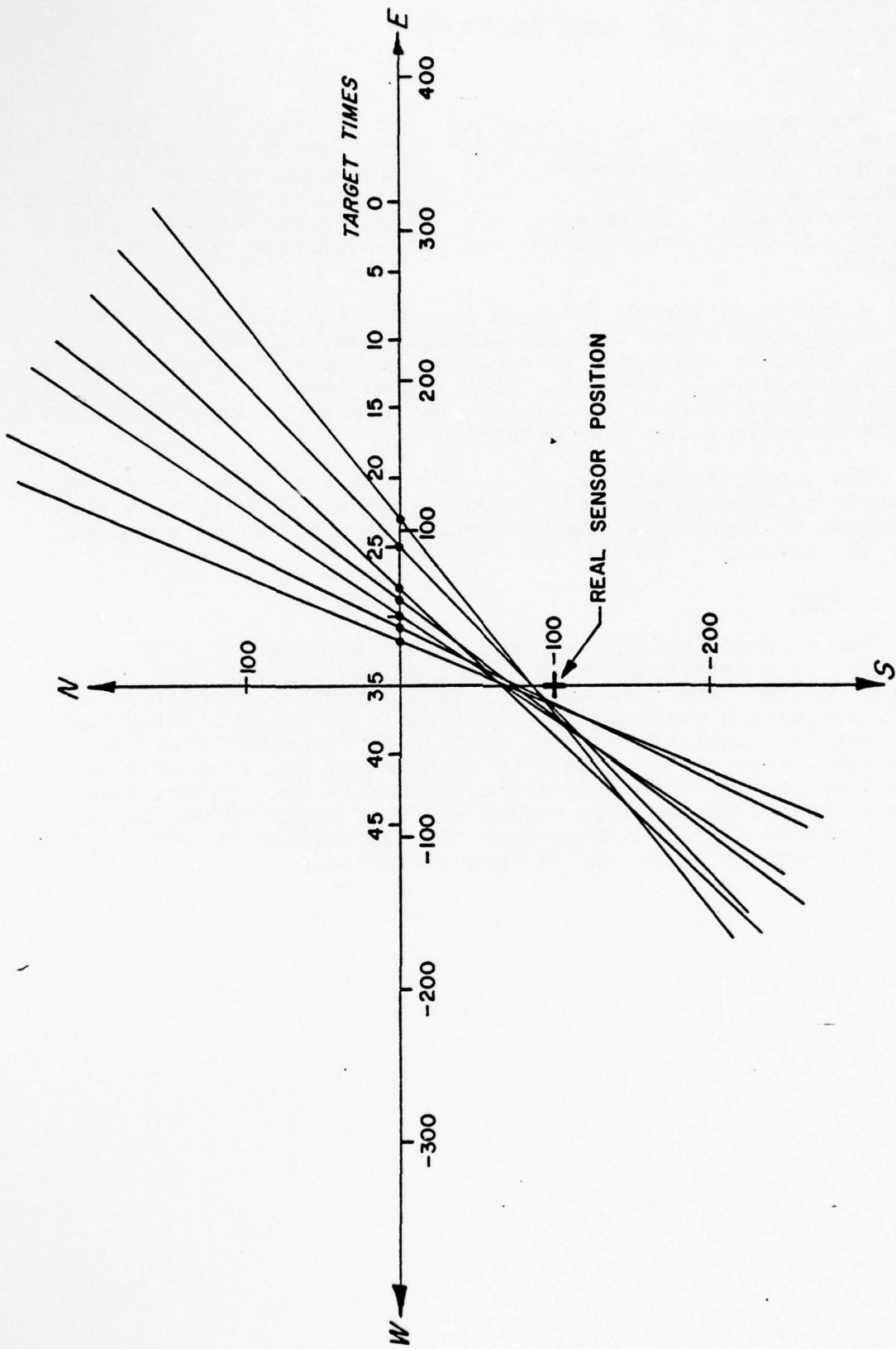
The reason the SPL calculation with the bias calculated was so poor was because the reports were few and very close together. A good bias calculation occurs when there are many reports and when the reports are widely spaced. The error occurs because the angle difference with close reports is often just a bearing bin shift and is not truly representative of bearing movement.

A graphical sensor location is shown in Figure 15 for the SPL calculation where the bias was set equal to zero. This was accomplished by drawing a line along the corrected report bearing from the vehicle location for each sensor report. The intersection of these lines locates the sensor position as is done arithmetically in the computer using this program.

The sensor location accuracy of these runs could be improved by applying appropriate weighting functions to the data, adding a compass to the sensor, or by running several target vehicles past the sensor.

Conclusions

For a sensor which reports a target bearing angle in addition to a detection, this algorithm can determine the location of that sensor after deployment. This report discusses the method used, presents a computer program capable of accomplishing the task, and does this analysis for two sensors of "unknown" location. Although better results could be obtained by applying weighting functions to the data or adding a compass to the sensor, this method located two sensors within 40 ft of their actual location. This location algorithm increases the flexibility of sensor placement for sensors which report target bearing.



SIMULATED SPL LINES (RUN #2); BIAS = 0 DEGREES

FIGURE 15

APPENDIX A
ERROR ANALYSIS

There are many factors which contribute to errors in accurately locating a sensor. These include:

1. Calculation of the rotational bias of the sensor, BIAS
2. Gaussian Noise, XK
3. Target course error, Δ ALA
4. Target X position error, Δ X
5. Target Y position error, Δ Y
6. Target Z position error, Δ Z
7. Target Velocity error, Δ V
8. Temperature error, Δ temp
9. The bearing resolution of the sensor, BINS

Errors due to these sources were computed for 58 simulated runs under four conditions. These conditions were:

1. Rotational bias calculated, one target vehicle
2. Rotational bias set, one target vehicle
3. Rotational bias calculated, two target vehicles
4. Rotational bias set, two target vehicles

The results of these runs are shown in Table 1. Unless specified the input parameters were:

BIAS = 0 degrees	V = 0 meters/sec
Δ XK = 0	temp = 0 degrees
Δ ALA = 0 degrees	number of Bins = 100,000 (runs 1-47)
Δ X = 0 meters	
Δ Y = 0 meters	number of Bins = 256 (runs 53-58)
Δ Z = 0 meters	

These sensor location errors were calculated for sensor positions 100, 500, and 1000 meters from the target track. The vehicle was traveling at 50 m/s at an altitude of 200 m. The range of the target track was 3000 meters. In Table 1, DR₁₀₀, DR₅₀₀, and

DR₁₀₀₀ refer to the distance of the sensor from the test track.

All sensor location errors are in meters displayed from their actual location.

Bias Errors

In Table 1 the bias errors only affected the cases where the bias was set. In those cases, the bias was set equal to that error amount and the sensor position error was calculated. In the cases where the bias was calculated, it was calculated correctly as all other parameters were "error free." Using this correctly calculated bias the sensor position was accurate. Figure 16 shows the effect of bias errors on the SPL algorithm. The results are about the same for one or two targets, but do depend on the distance of the sensor from the target track.

Gaussian Noise

The signal received by the sensor from some target source contains noise. This noise affects the bearing angle reported by the sensor.

All noise is assumed to be Gaussian added to the real report bearing. This noise consists of an average value (Mean) and a standard deviation (Sigma). The mean is just the bias. Sigma is a function of the signal to noise ratio. The higher the ratio the lower the error. The following relationship has been observed:

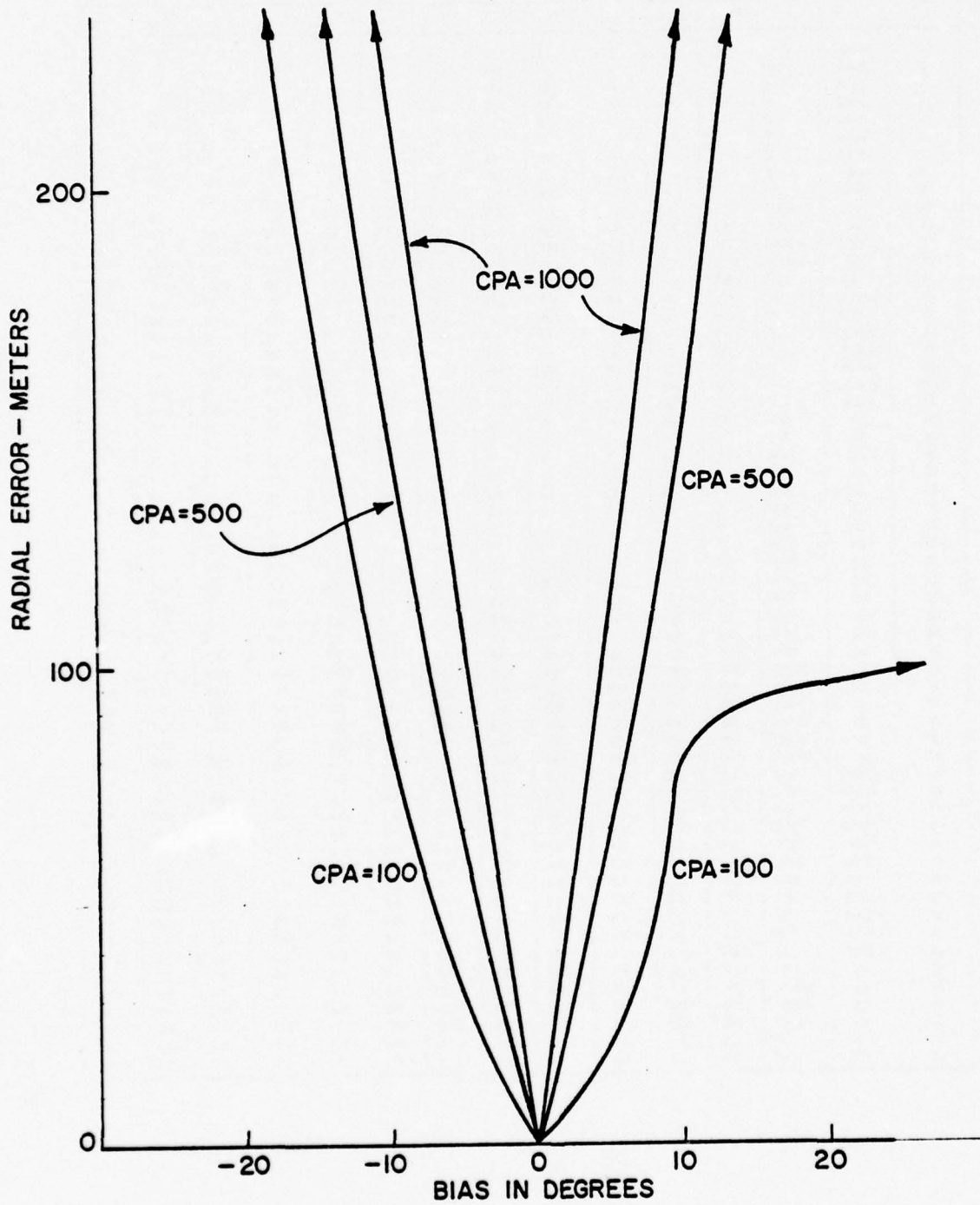
$$\text{Sigma} = \frac{K_1}{S/N} \quad (35)$$

S/N is inversely proportional to the radial distance from the sensor to the target. Equation 36 depicts this relationship. where S/N_{cpa} is the signal to noise ratio at the closest point of approach (cpa).

TABLE 1 ERROR ANALYSIS OF SENSOR POSITION LOCATER

RUN NUMBER	ERROR TYPE	ERROR AMOUNT	ONE TARGET						TWO TARGETS					
			WITH CALCULATED BIAS			WITH BIAS SET			WITH CALCULATED BIAS			WITH BIAS SET		
			DR ₁₀₀	DR ₅₀₀	DR ₁₀₀₀	DR ₁₀₀	DR ₅₀₀	DR ₁₀₀₀	DR ₁₀₀	DR ₅₀₀	DR ₁₀₀₀	DR ₁₀₀	DR ₅₀₀	DR ₁₀₀₀
1	NONE	-	0	0	0	0	0	0	0	0	0	0	0	0
2	BIAS	20°	-	-	-	96	385	517	-	-	-	282	347	515
3	BIAS	10°	-	-	-	79	145	229	-	-	-	68	141	258
4	BIAS	5°	-	-	-	21	63	108	-	-	-	23	64	129
5	BIAS	1°	-	-	-	3	12	21	-	-	-	2.6	12	26
6	BIAS	1°	-	-	-	3	12	21	-	-	-	2.5	12	26
7	BIAS	5°	-	-	-	26	65	107	-	-	-	17	62	126
8	BIAS	10°	-	-	-	82	149	225	-	-	-	50	128	241
9	BIAS	20°	-	-	-	299	362	519	-	-	-	463	342	487
10	XK	0.0001	0.2	0.8	2.4	0.1	0.5	1.5	0.2	0.7	0.3	0.1	0.3	1.1
11	XK	0.0010	12.1	9.1	28.9	11.6	6.5	16.8	7.6	9.3	35	4.5	9.0	16.4
12	XK	0.0050	416	216	509	265	150	363	127	32	203	337	33	85
13	XK	0.0100	892	2074	901	120	1904	4023	1395	1244	289	476	2154	876
14	ΔALA	20°	869	885	935	960	546	1263	901	1038	1236	2961	1301	1565
15	ΔALA	10°	436	444	469	514	349	560	444	447	491	521	505	549
16	ΔALA	5°	218	222	235	235	199	255	209	177	188	212	175	154
17	ΔALA	1°	43	44	47	44	44	47	34	29	32	32	18	8.8
18	ΔALA	1°	43	44	47	42	46	45	38	30	33	37	18	9.3
19	ΔALA	5°	218	222	235	195	248	205	210	178	189	228	181	159
20	ΔALA	10°	436	444	469	355	538	358	403	435	445	503	636	489
21	ΔALA	20°	869	885	935	899	1190	546	884	1020	1040	504	1171	1332
22	ΔX	10 M	10	10	10	10	10	10	10	10	10	10	10	10
23	ΔX	5 M	5	5	5	5	5	5	5	5	5	5	5	5
24	ΔX	5 M	5	5	5	5	5	5	5	5	5	5	5	5
25	ΔX	10 M	10	10	10	10	10	10	10	10	10	10	10	10
26	ΔY	10 M	10	10	10	10	10	10	10	10	10	10	10	10
27	ΔY	5 M	5	5	5	5	5	5	5	5	5	5	5	5
28	ΔY	5 M	5	5	5	5	5	5	5	5	5	5	5	5
29	ΔY	10 M	10	10	10	10	10	10	10	10	10	10	10	10
30	ΔZ	100 M	13.4	6.5	4.0	13.4	6.0	3.4	15.0	6.1	3.9	14.9	5.9	3.7
31	ΔZ	50 M	6.6	3.0	1.8	6.6	2.8	1.5	6.6	2.8	1.8	6.6	2.7	1.7
32	ΔZ	10 M	1.3	0.5	0.3	1.3	0.5	0.3	1.2	0.5	0.3	1.2	0.5	0.3
33	ΔZ	10 M	1.3	0.5	0.3	1.3	0.5	0.3	1.2	0.5	0.3	1.2	0.5	0.3
34	ΔZ	50 M	6.2	2.4	1.4	6.1	2.2	1.2	5.3	2.2	1.4	5.3	2.2	1.3
35	ΔZ	100 M	11.6	4.1	2.4	11.5	3.8	2.1	9.6	3.7	2.4	9.5	3.7	2.2
36	ΔV	20 M/S	995	987	1027	994	975	995	1035	1186	1410	1034	1173	1378
37	ΔV	10 M/S	498	495	517	498	490	503	516	602	741	515	597	733
38	ΔV	5 M/S	249	248	259	249	246	252	252	307	390	252	305	389
39	ΔV	5 M/S	249	249	261	249	248	256	260	301	385	290	301	385
40	ΔV	10 M/S	499	500	523	499	497	514	516	856	749	516	612	751
41	ΔV	20 M/S	999	1005	1054	999	1000	1040	1038	1210	1458	1039	1205	2635
42	ΔTEMP	20°C	1.2	2.6	4.0	1.3	2.3	6.0	1.1	3.0	6.1	1.1	3.2	6.4
43	ΔTEMP	10°C	0.6	1.3	2.0	0.7	1.7	3.1	0.6	1.5	3.1	0.6	1.6	3.3
44	ΔTEMP	5°C	0.3	0.7	1.0	0.3	0.9	1.5	0.3	0.8	1.6	0.3	0.8	1.7
45	ΔTEMP	5°C	0.3	0.7	1.0	0.3	0.9	1.6	0.3	0.8	1.6	0.3	0.8	1.7
46	ΔTEMP	10°C	0.6	1.4	2.1	0.7	1.8	3.2	0.6	1.6	3.2	0.6	1.7	3.4
47	ΔTEMP	20°C	1.3	2.8	4.3	1.4	3.6	6.5	1.2	3.2	6.5	1.2	3.4	6.9
48	# BINS	1024	0.2	0.2	1.3	0.1	0.2	1.3	0.1	0.1	0.3	0.0	0.2	0.2
49	# BINS	512	0.6	1.9	2.6	0.4	1.8	2.0	0.3	1.0	0.5	0.3	0.6	0.3
50	# BINS	256	0.6	3.7	2.8	0.4	0.1	2.5	0.6	2.3	2.3	0.2	0.0	0.5
51	# BINS	128	0.6	5.7	18.4	0.1	5.0	11.6	0.6	3.2	10.3	0.1	1.2	2.1
52	# BINS	64	5.4	23.4	12.9	3.1	22.7	26.0	2.6	6.3	24.3	1.8	5.3	23.1
53	XK	0.0001	0.2	6.0	9.4	0.3	2.5	5.2	0.3	1.3	4.6	0.2	1.8	0.9
54	XK	0.0010	7.5	9.3	32	6.4	5.7	32	4.4	6.4	16	3.0	5.9	10.6
55	XK	0.0050	623	275	690	146	228	636	300	148	123	413	116	114
56	XK	0.0100	1670	505	5284	5687	689	2266	4261	2763	203	710	352	101
57	BIAS	5°	1.0	5.2	4.0	239	63.7	107.5	0.4	2.1	1.9	22.5	63.4	128
58	BIAS	1°	0.4	10.2	5.0	37	12.7	21.2	0.6	2.4	4.6	2.7	12.3	25.7

AFFECTS OF BIAS ON SPL ERRORS



$$S/N = f(D) = \frac{K_2 + S/N_{cpa}}{\sqrt{D + 1}} \quad (36)$$

Combining equations 35 and 36 gives:

$$\text{Sigma} = K \sqrt{D + 1} \quad (37)$$

For the error analysis values of .0001, .001, .005, and .01 were used for K. Figure 17 plots the radial error (meters) of the sensor as a function of K. From Figure 17 it is noted that repeating the calculations for a second target greatly reduces the error from this source.

Target Course Error

This error comes about when the test vehicle is traveling in a direction slightly different than it is thought to be traveling. Figure 18 plots these errors as a function of course error (in degrees).

ΔX , ΔY Errors

This type of error occurs when the target vehicle is at a point different than it is thought to be. Equation 38 gives the radial error of the sensor as a function of the ΔX , ΔY displacement.

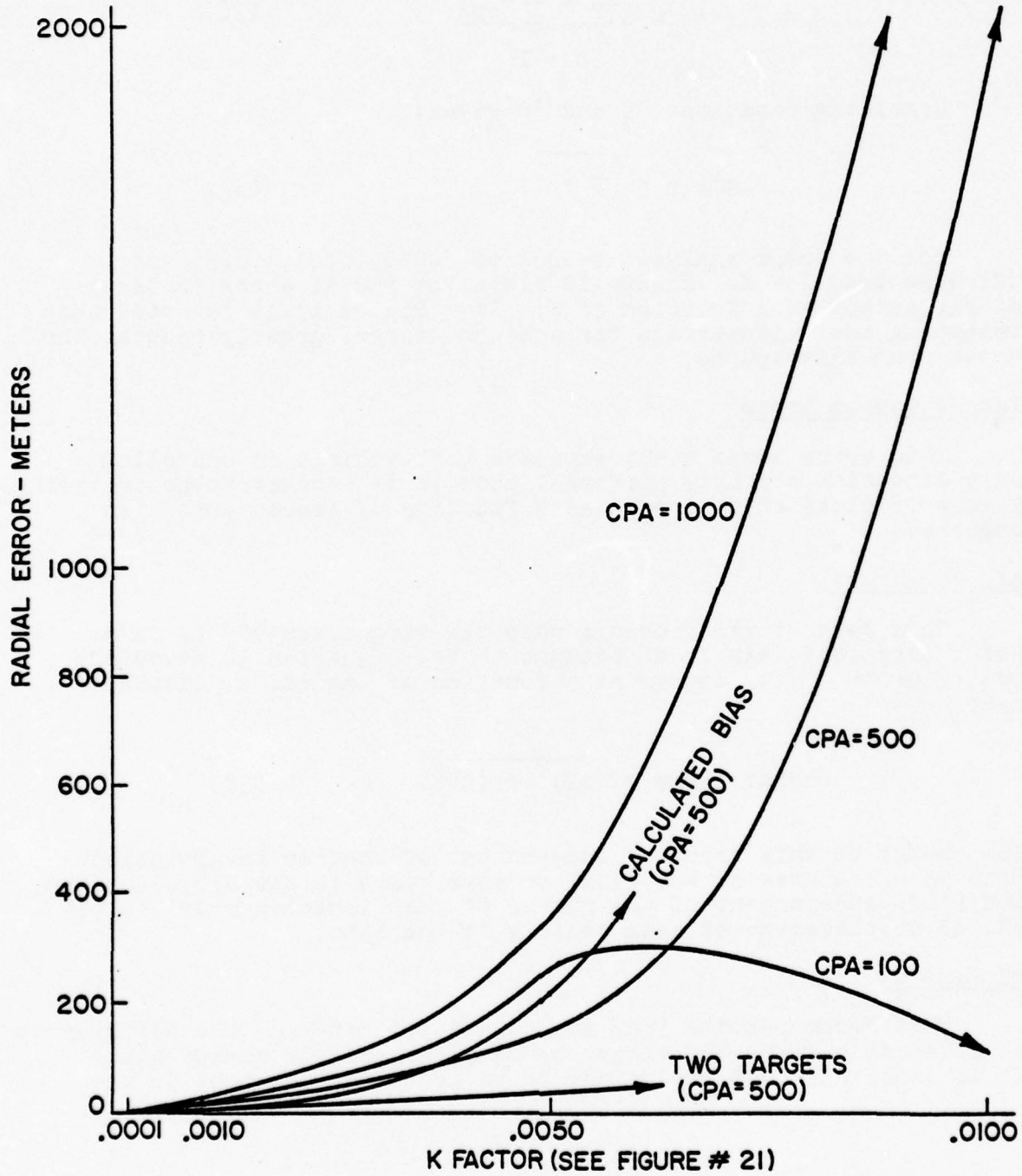
$$\text{radial error} = \sqrt{(\Delta X)^2 + (\Delta Y)^2} \quad (38)$$

The amount of this error is independent of whether the rotational bias is calculated or set equal to some value in the SPL algorithm, and it is independent of the number of test vehicles provided the ΔX , ΔY displacement of each vehicle is the same.

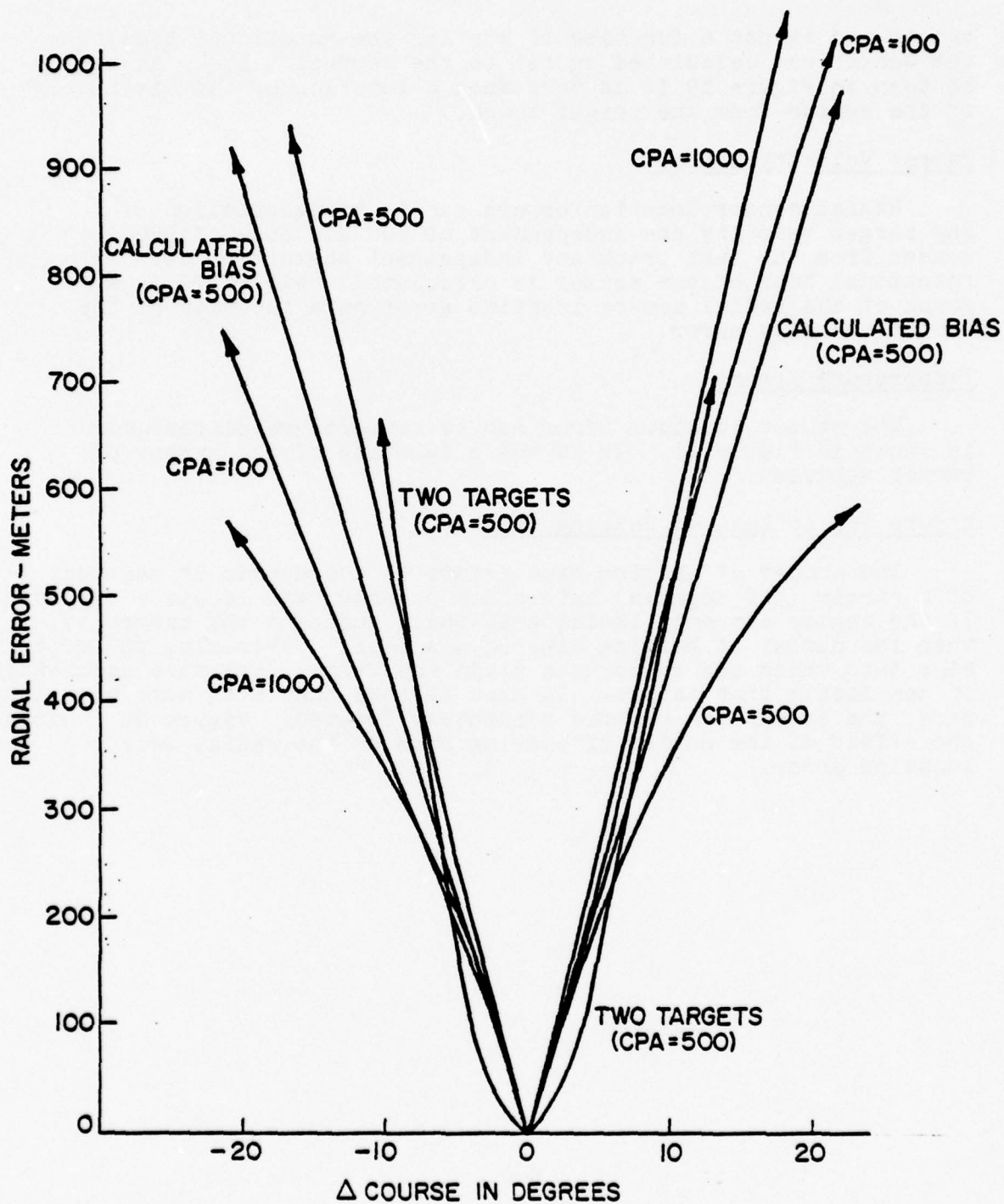
ΔZ Errors

This error results from inaccurate estimates of the differences in altitude between the target track plane and the sensor plane. It is independent of the number of targets used to sight in the

AFFECTS OF K (GAUSSIAN NOISE) ON SPL ERRORS



AFFECTS OF Δ COURSE ON SPL ERRORS



sensor and is not a function of whether the rotational bias of the sensor was calculated or set to the correct value. As can be seen in Figure 19 it is very much a function of the distance of the sensor from the target track.

Target Velocity Errors

Radial sensor location errors due to miscalculation of the target velocity are independent of the distance of the sensor from the test track and independent whether or not the rotational bias of the sensor is calculated. Figure 20 is a graph of the radial sensor location error as a function of the target velocity error.

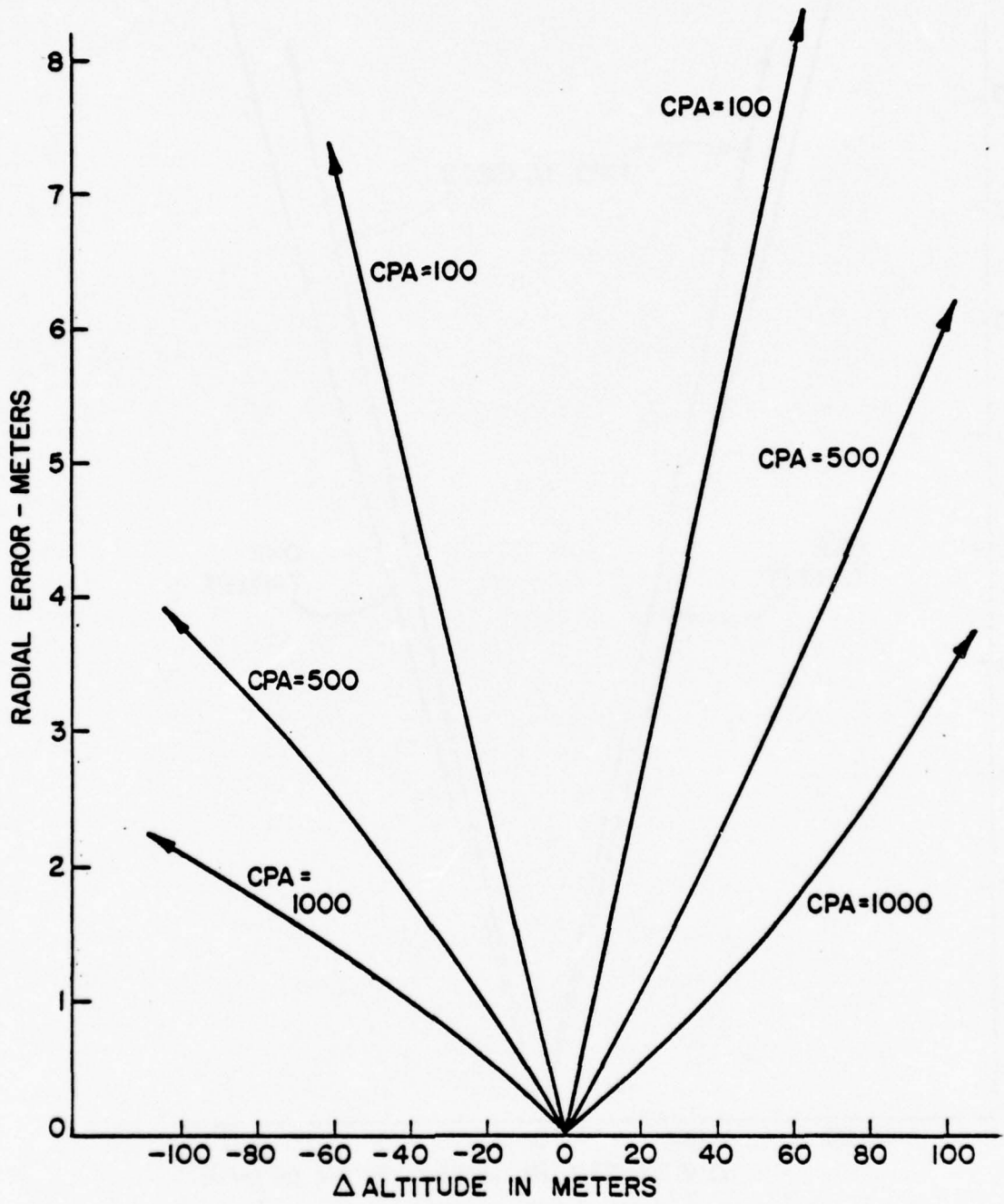
Temperature Errors

The sensor location error due to temperature differences is shown in Figure 21. It is not a function of the number of target vehicles.

Errors Due to Lack of Bearing Bins

The number of bearing bins refers to the number of sections of a circle (360 degrees) into which a sensor can locate a target. If the sensor can only indicate in which quadrant the target is, then the number of bearing bins equals four. Obviously, the more bins into which the sensor can place the target, the more accurately it can locate that target. It also follows that with more bearing bins, the sensor can be more accurately located. Figure 22 depicts the affect of the number of bearing bins on the radial sensor location error.

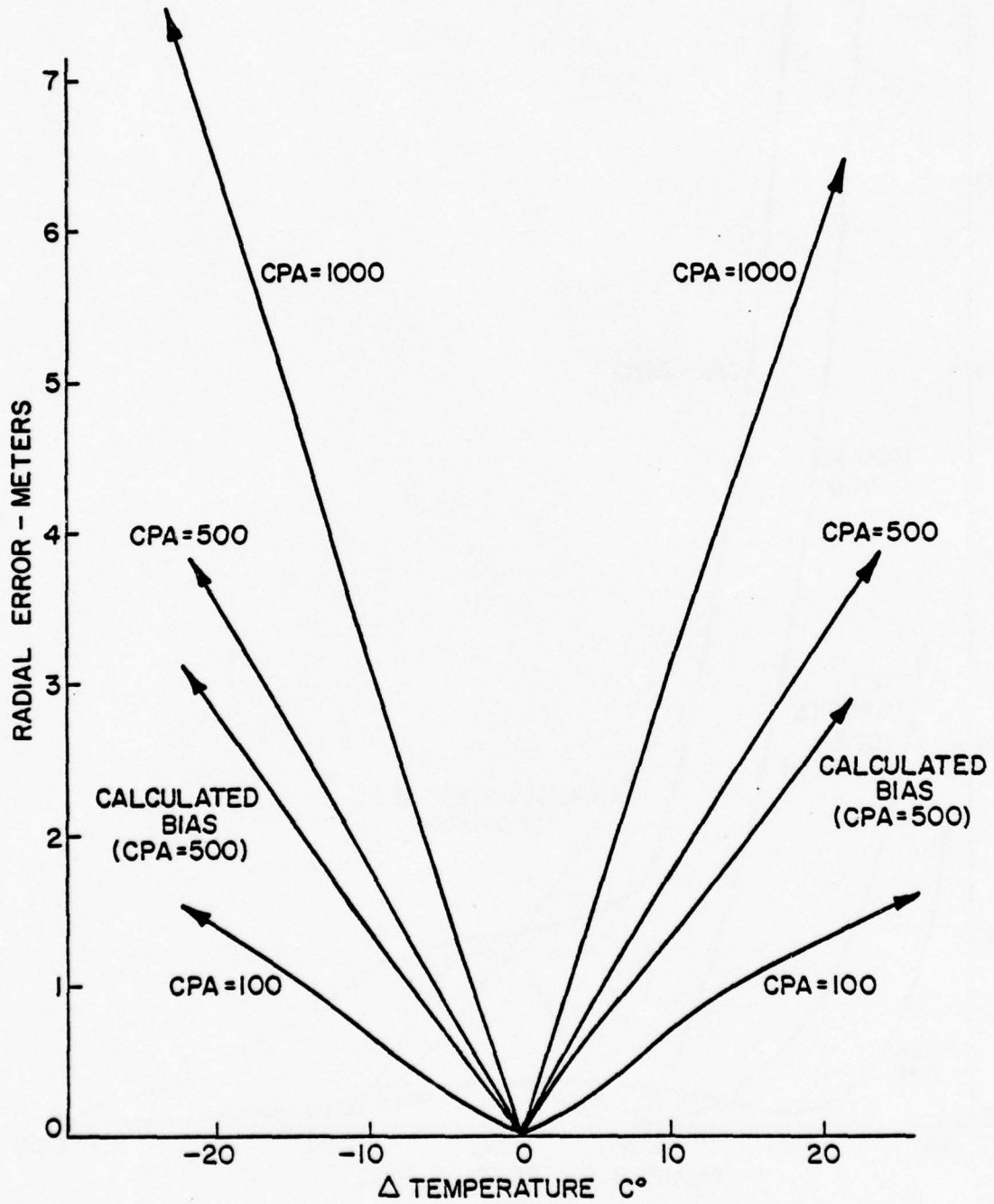
AFFECTS OF ΔZ ON SPL ERRORS



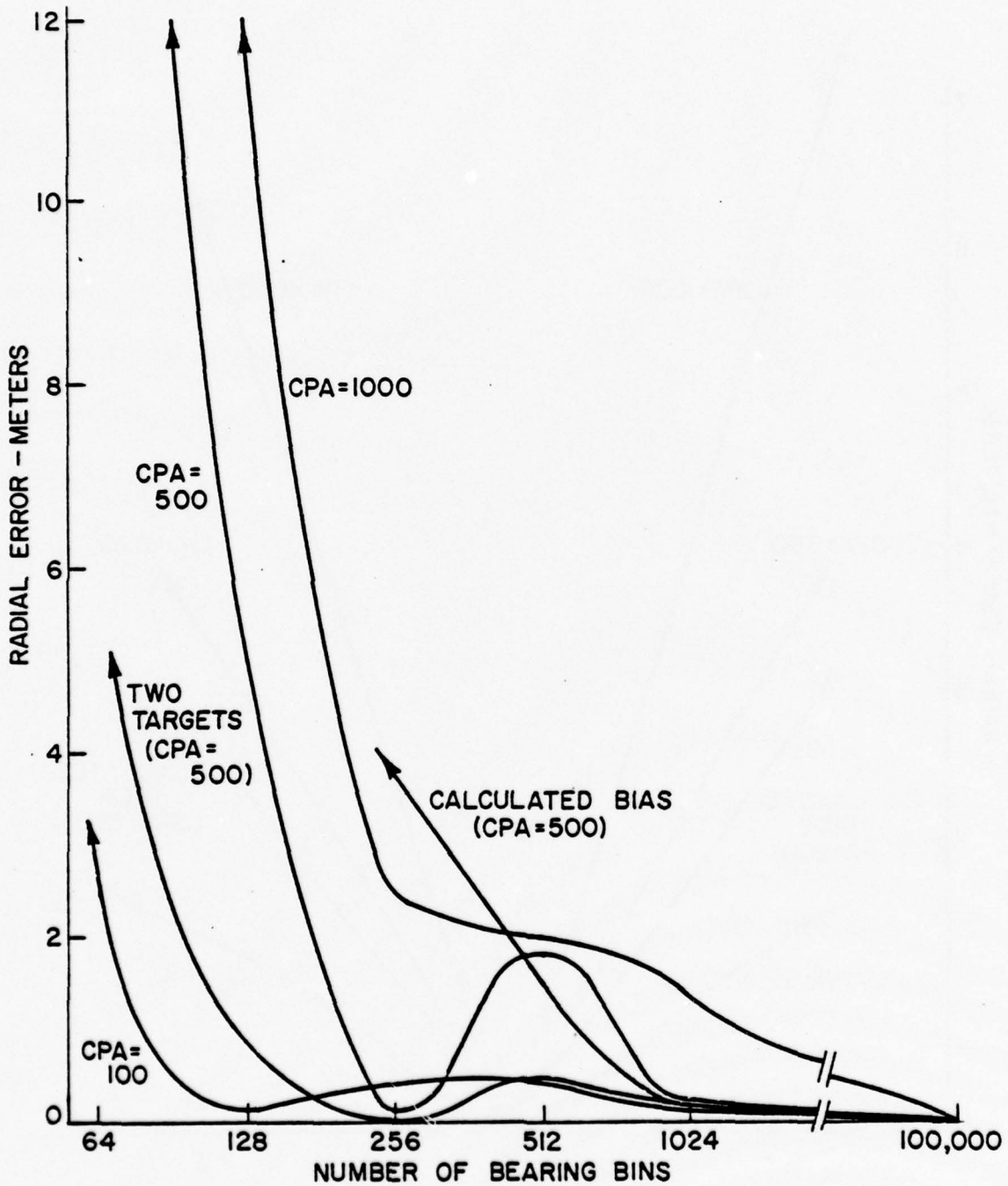
AFFECTS OF Δ VELOCITY ON SPL ERRORS



AFFECTS OF Δ TEMPERATURE ON SPL ERRORS



AFFECTS OF NUMBER OF BEARING BINS ON SPL ERRORS



APPENDIX B

FORTRAN PROGRAMS AND FLOW CHARTS

Since there are many calculations involved in locating a sensor position, it is best done using a computer. Appendix B is a listing of the flow charts and the computer program used to accomplish this. The computer program is called Simulate, and it calls nine subroutines called Input, Senchk, Output, SPL, Enter, Shift, TXY, Normal and Uniform.

Program Simulate is the program which locates the sensor position. It accomplishes this by calling four subroutines; INPUT, SENCHK, SPL, and OUTPUT.

Subroutine INPUT loads the initial parameters into the program.

Subroutine SENCHK picks which sensor reports should be used in the calculation. For those reports it provides the time of report, report bearing, and a weighting function to the main program.

Subroutine SPL calculates the sensor location and the rotational bias from the data provided to it from the main program.

Subroutine OUTPUT prints out the results of the calculations.

There are five other subroutines which are used in this program. Subroutines Normal and Uniform together provide a random number uniformly distributed between 0 and 1 to Subroutine SENCHK. Subroutines Enter, Shift, and TXY simply perform minor arithmetic calculations needed in the SPL Subroutine.

The flow charts, Figures 23-30, and computer printout for these programs, Figure 31, follow.

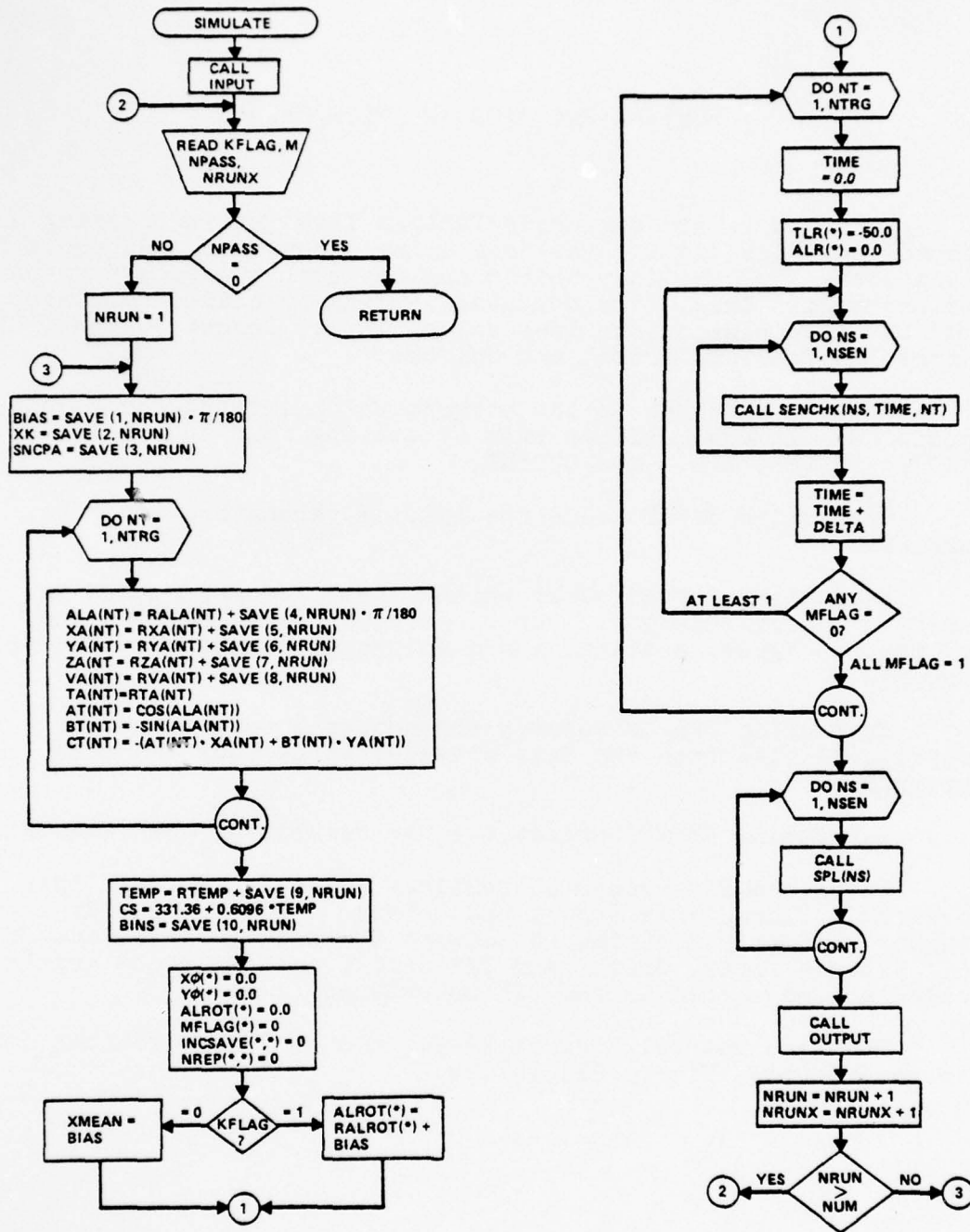


FIG. 23 FLOWCHART OF SIMULATE

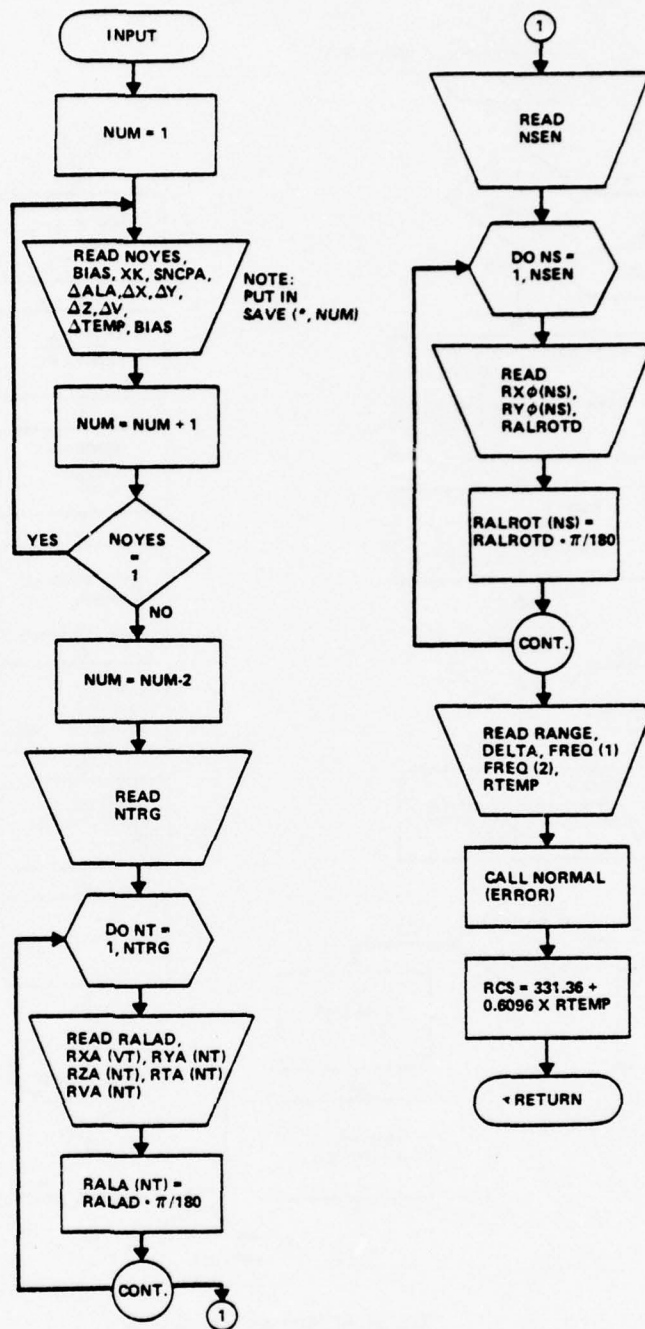


FIG. 24 FLOWCHART OF INPUT

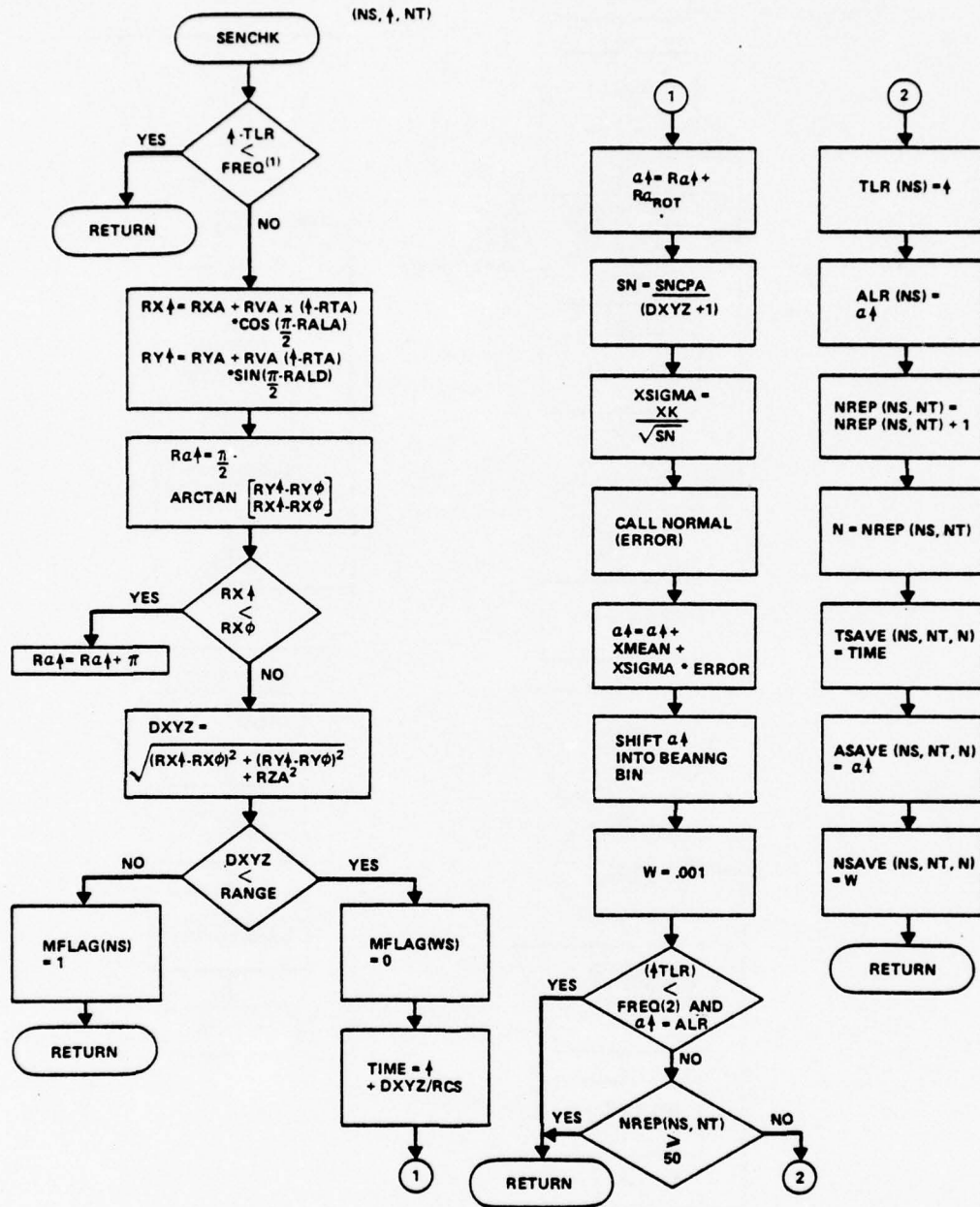


FIG. 25 FLOWCHART OF SENCHK

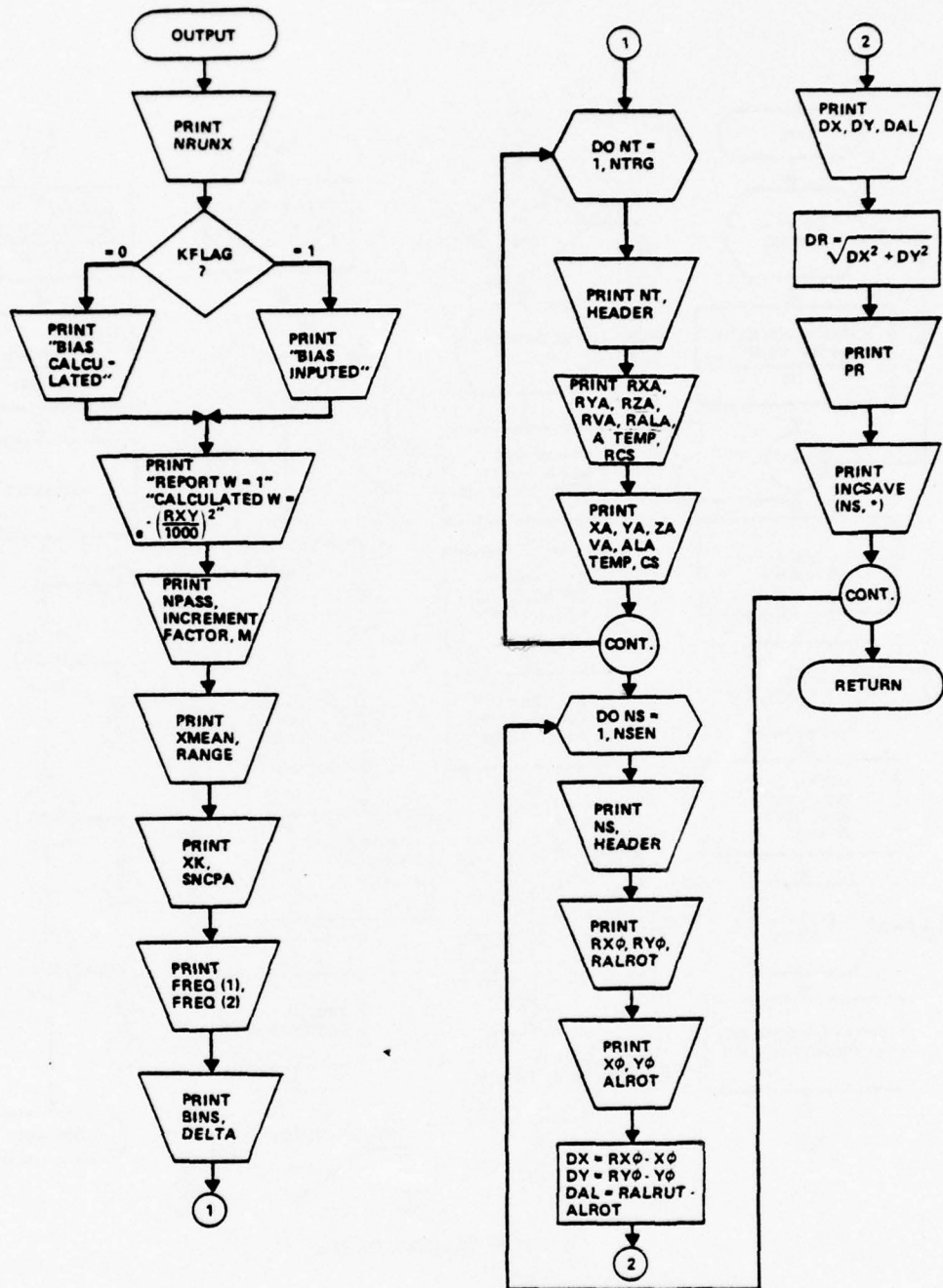


FIG. 26 FLOWCHART OF OUTPUT

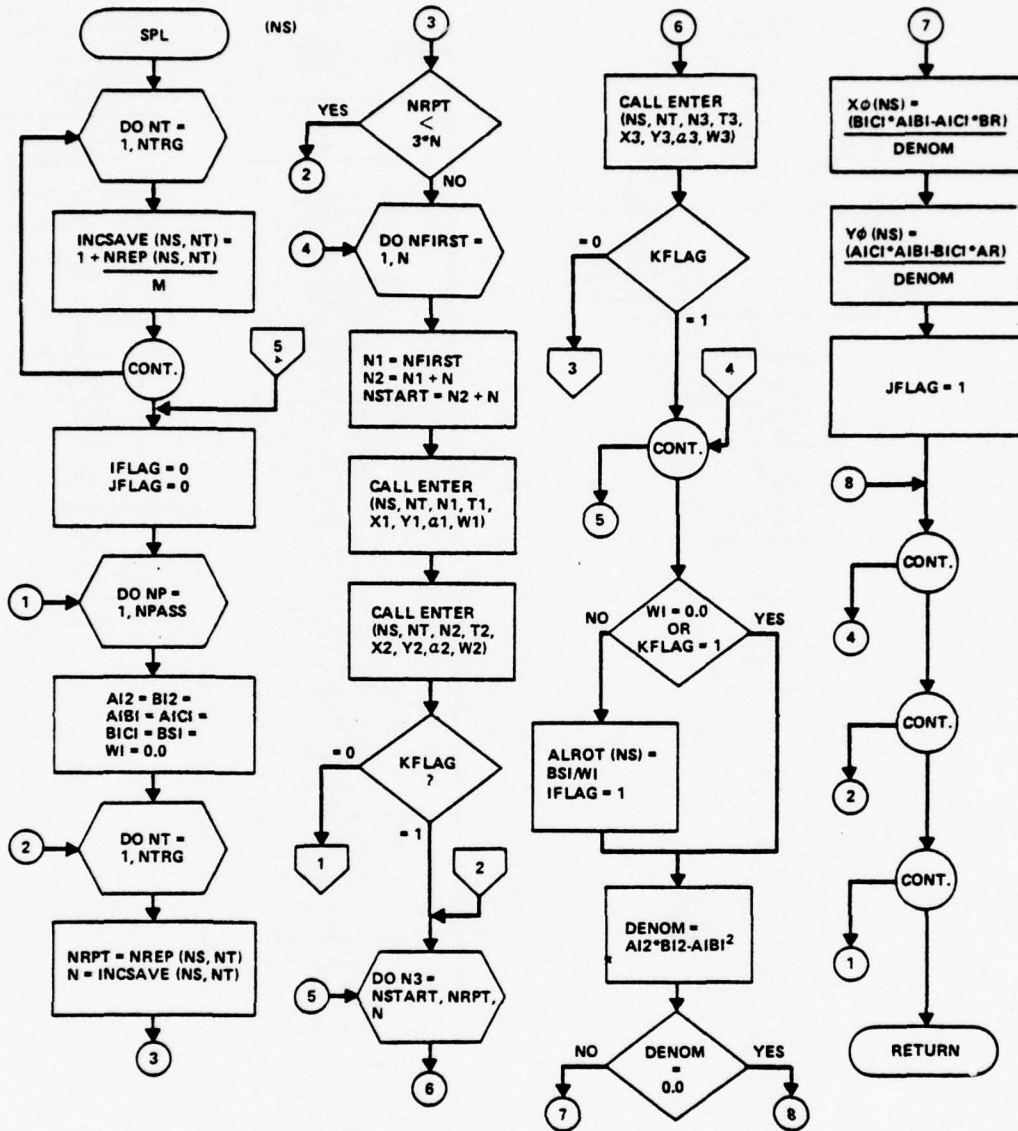


FIG. 27a FLOWCHART OF SPL

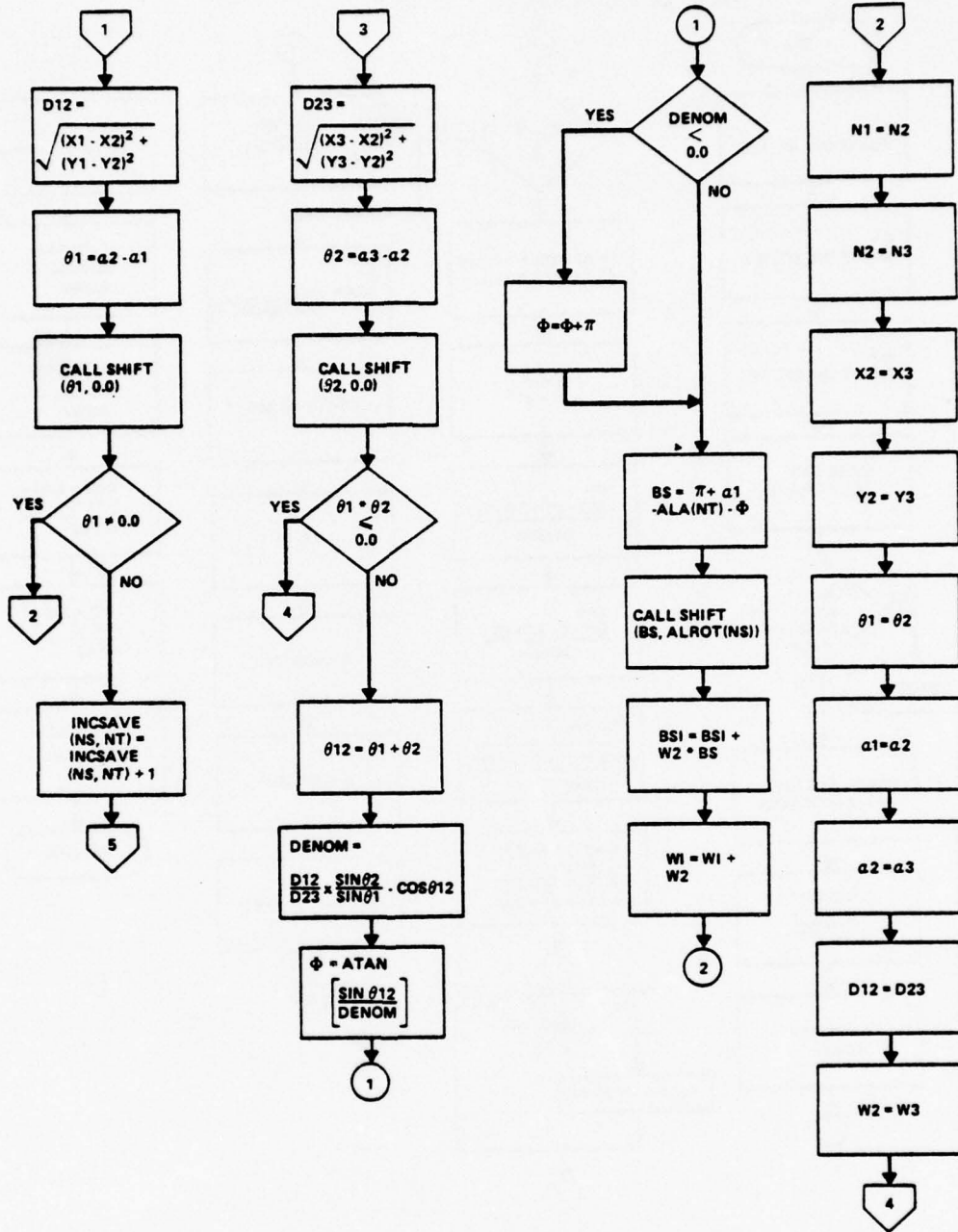


FIG. 27b FLOWCHART OF SPL (CONT.)

(NS, NT, NR, T, XT, YT, a, W)

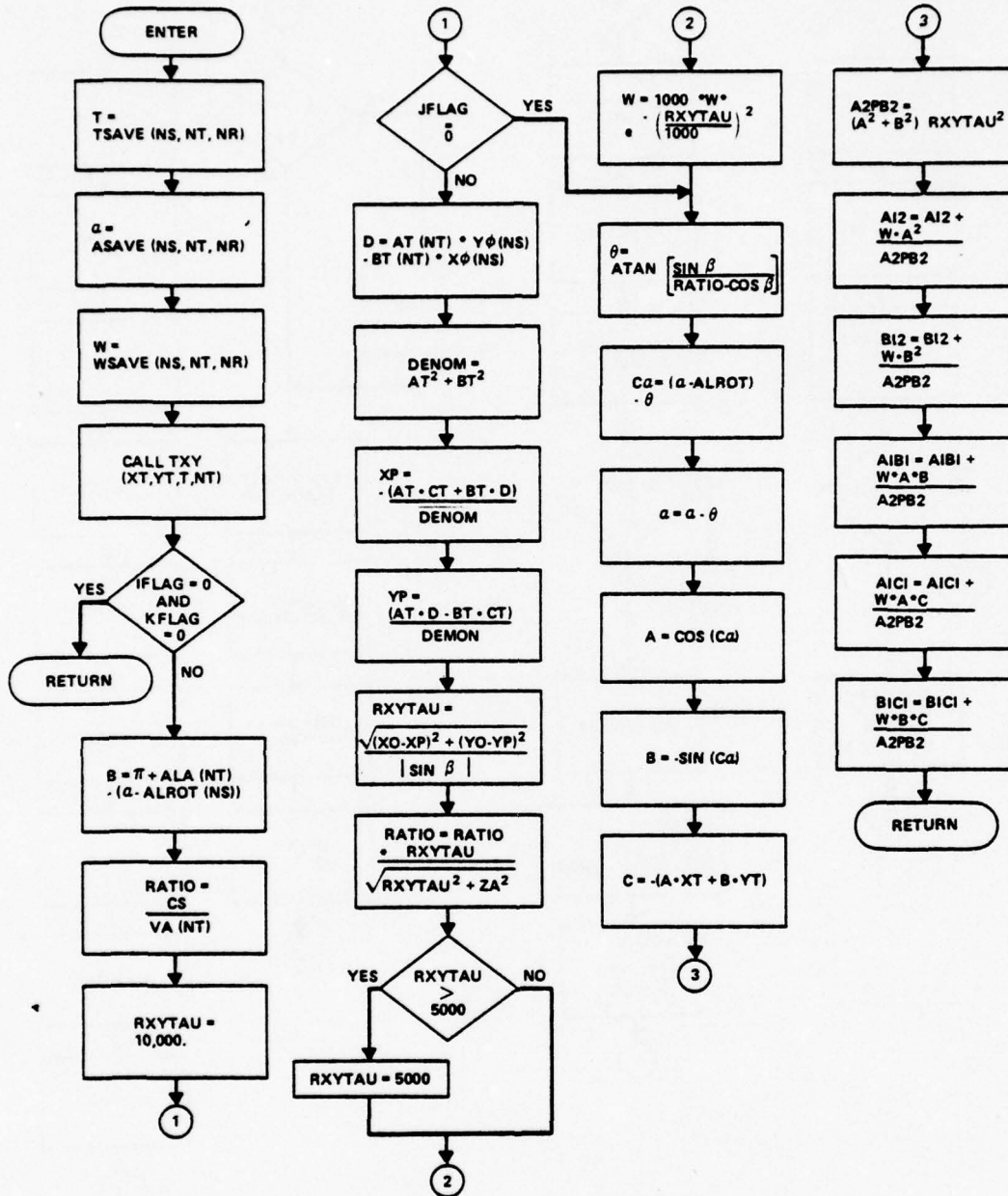


FIG. 28 FLOWCHART OF ENTER

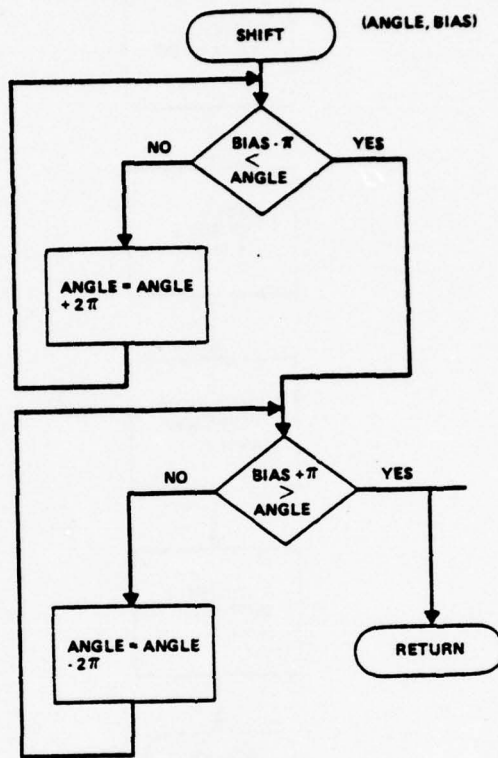


FIG. 29 FLOWCHART OF SHIFT

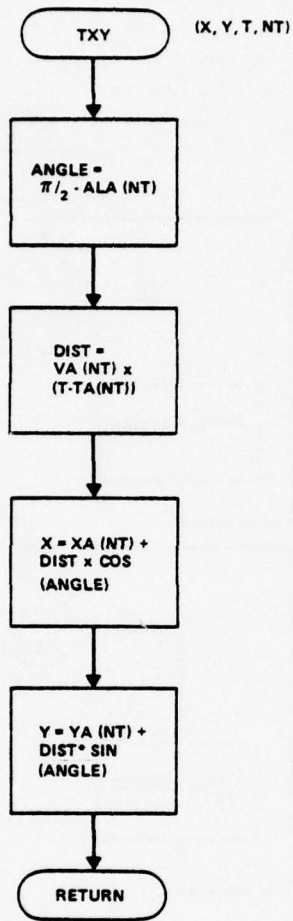


FIG. 30 FLOWCHART OF TXY

00000

PROGRAM SIMULATE

```

C
C
C .....
C *
C * JONATHAN VALVANO 03/18/75 *
C *
C * THESE PROGRAMS SIMULATE A SENSOR FIELD FOR THE
C * DEVELOPMENT OF A SENSOR POSITION LOCATOR (SPL).
C * ACOUSTIC SENSORS ARE DEPLOYED IN THE FIELD WITHOUT
C * THE KNOWLEDGE OF THEIR EXACT LOCATION. THIS CAN BE DONE
C * BY AIR DROP OR GUN DELIVERY. THE PURPOSE OF THESE
C * ROUTINES IS TO LOCATE THE SENSORS ONCE THEY ARE IN THE
C * FIELD. THE FOLLOWING METHOD IS USED....
C * 1. A TARGET, WHOSE POSITION AS A FUNCTION OF TIME
C * IS KNOWN, IS RUN OVER THE SENSOR FIELD.
C * 2. THE RECEIVING UNIT LISTENS AND RECORDS THE SEQUENCE
C * OF REPORTS BY THE SENSOR OF THE TARGET.
C * 3. THE ARRAYS OF REPORTS PLUS THE TARGET LOCATIONS
C * ARE FED AS INPUTS TO THE SPL ALGORITHM.
C * 4. SPL CALCULATES THE SENSOR LOCATION (AND BIAS ANGLE)
C *
C * THE FOLLOWING OPTIONS ARE AVAILABLE TO SPL...
C * KFLAG=0 IF THE BIAS IS TO BE CALCULATED.
C * KFLAG=1 IF THE BIAS IS REPORTED ( INPUTED TO SPL.)
C * M IS THE INCREMENT FACTOR (SEE SPL)
C * NPASS IS THE NUMBER OF PASSES SPL MAKES OVER REPORTS
C *
C * THE FOLLOWING OPTIONS ARE AVAILABLE TO THE SIMULATION...
C * NSEN THE NUMBER OF SENSORS
C * NTRG THE NUMBER OF TARGETS
C * RANGE THE RANGE OF THE SENSORS
C * BINS THE NUMBER OF BEARING BINS OF THE SENSOR
C * FREQ(1) THE CYCLE TIME WHEN THE BEARING CHANGES
C * FREQ(2) THE CYCLE TIME WHEN THE BEARING IS THE SAME
C * DELTA THE TIME INCREMENT IN THE SIMULATION
C * XMEAN,XK,SINCPA ARE FOR GAUSSIAN ERROR
C * MEAN=XMEAN
C * SIGMA=XK/SQRT(SINCPA/(DXYZ+1))
C *
C .....

```

```

00001 COMMON /TRGT// ALA(3),XA(3),YA(3),ZA(3),
00001 1 TA(3),VA(3),AT(3),BT(3),CT(3),TEMP,CS,NTRG
00002 COMMON /RTGT// RALA(3),RXA(3),RYA(3),RZA(3),
00002 1 RTA(3),RVA(3),RTEMP,PCS
00003 COMMON /SEN// NSEN,X0(3),Y0(3),ALROT(3),IFLAG,JFLAG
00004 COMMON /HSEN// RX0(3),RY0(3),RALROT(3)
00005 COMMON /REP// TSAVE(3,3,50),ASAVE(3,3,50),WSAVE(3,3,50),
00005 1 NREP(3,3),INCSAVE(3,3)
00006 COMMON /STATE// TLR(3),ALP(3),MFLAG(3),FREQ(2),
00006 1 HINS,XMEAN,XK,SINCPA,RANGE,DELTA,NUM,NHUN,NRUNK,SAVE(10,100)
00007 COMMON /CONST// PI,PI02,PI2
00008 COMMON /OPTIONS// KFLAG,M,NPASS

```

FIG. 31 COMPUTER PROGRAM

```

C
C .....
C *
C *   FIRST CALL INPUT TO SET UP THE SIMULATION
C *
C .....
00009 C       CALL INPUT
C
C .....
C *
C *   THEN READ THE OPTIONS FOR THIS SET OF RUNS
C *   THIS IS THE START OF A SET OF RUNS WITH SEPERATE OPTIONS.
C *
C .....
00010 C       101  READ 103, KFLAG, M, NPASS, NRUNX
00011 C       103  FORMAT (4I10)
00012 C           IF (NPASS .EQ. 0) RETURN
00014 C           NRUN=1
C
C .....
C *
C *   SAVE CONTAINS THE SIMULATION OPTIONS FOR EACH RUN OF SET
C *
C *   ALL ANGLE INPUTS AND OUTPUTS ARE IN DEGREES AND
C *   ALL INTERNAL ANGLES ARE IN RADIAN.
C *
C *   ALA IS THE INPUTED TARGET COURSE.
C *   RALA IS THE REAL   TARGET COURSE.
C *   VA IS THE INPUTED TARGET VELOCITY.
C *   RVA IS THE REAL   TARGET VELOCITY.
C *   AT TIME TA (SECONDS THE TARGET IS AT (XA,YA,ZA) INPUTED.
C *   AT TIME RTA (SEC) THE TARGET IS AT (RXA,RYA,RZA) REAL.
C *
C .....
00015 C       110  BIAS=SAVE(1,NRUN)*PI/180.0
00016 C           XK=SAVE(2,NRUN)
00017 C           SNCPA=SAVE(3,NRUN)
00018 C           DO 155 NT=1,NTRG
00019 C           ALA(NT)=RALA(NT)+SAVE(4,NRUN)*PI/180.0
00020 C           XA(NT)=RXA(NT)+SAVE(5,NRUN)
00021 C           YA(NT)=RYA(NT)+SAVE(6,NRUN)
00022 C           ZA(NT)=RZA(NT)+SAVE(7,NRUN)
00023 C           VA(NT)=RVA(NT)+SAVE(8,NRUN)
00024 C           TA(NT)=RTA(NT)
C
C
C
C

```

```

C
C .....
C *
C *   AT,BT,CT FORM THE LINE AT*X+BT*Y+CT=0 OF THE TARGET TRACK.
C *
C * .....
C
00025      AT(NT)=COS(ALA(NT))
00026      BT(NT)=-SIN(ALA(NT))
00027      CT(NT)=- (AT(NT)*XA(NT)+BT(NT)*YA(NT))
00028      155 CONTINUE

C
C .....
C *
C *   CS IS THE INPUTED SPEED OF SOUND IN METERS/SECOND.
C *   RCS IS THE REAL   SPEED OF SOUND IN METERS/SECOND.
C *   TEMP IS THE INPUTED TEMPERATURE IN CENTIGRADE.
C *   RTEMP IS THE REAL   TEMPERATURE IN CENTIGRADE.
C *
C * .....
00029      TEMP=RTEMP+SAVE(9,NRUN)
00030      CS=331.36+0.6096*TEMP
00031      BINS=SAVE(10,NRUN)
00032      DO 160 I=1,3

C
C .....
C *
C *   (X0,Y0) IS THE CALCULATED SENSOR POSITION.
C *   (RX0,RY0) IS THE REAL SENSOR POSITION.
C *   ALROT IS EITHER THE CALCULATED OR INPUTED SENSOR
C *   ANGLE ROTATION BIAS, DEPENDING ON OPTION= KFLAG.
C *   RALROT IS THE REAL SENSOR ANGLE ROTATION BIAS.
C *
C *   MFLAG=0 WHEN THE TARGET IS WITHIN RANGE OF THAT SENSOR.
C *   MFLAG=1 WHEN THE TARGET IS OUTSIDE THE RANGE OF THAT SENSOR.
C *
C * .....
00033      X0(I)=0.0
00034      Y0(I)=0.0
00035      ALROT(I)=0.0
00036      MFLAG(I)=0
00037      IF (KFLAG .EQ. 0) XMEAN=BIAS
00039      IF (KFLAG .EQ. 1) ALROT(I)=RALROT(I)+BIAS
00041      DO 160 J=1,3
00042      INCSAVE(I,J)=0
00043      160 NREP(I,J)=0

```

```

00044      DO 125 NT=1,NTRG
00045      TIME=0.0
00046      DO 111 I=1.3
C
C
C *****
C *
C *   TLR IS THE TIME OF THE LAST REPORT.
C *   ALR IS THE ANGLE OF THE LAST REPORT.
C *
C *****
C
00047      TLR(I)=-50.0
00048      ALR(I)=0.0
00049      111 CONTINUE
00050      115 DO 120 NS=1,NSEN
C
C *****
C *
C *   SENCHK WILL CHECK TO SEE IF THE SENSOR SHOULD REPORT.
C *   IF SO IT DETERMINES THE TIME OF THE REPORT AND PUTS IT
C *   IN TSAVE. IT DETERMINES THE HEARING AND PUTS IT IN ASAVE.
C *   IF THE SENSOR REPORTS A WEIGHTING FACTOR IT IS IN WSAVE.
C *
C *****
C
00051      120 CALL SENCHK(NS,TIME,NT)
00052      TIME=TIME+DELTA
00053      DO 121 NS=1,NSEN
00054      IF (MFLAG(NS) .EQ. 0) GO TO 115
00055      121 CONTINUE
00056      125 CONTINUE
C
C *****
C *
C *   NOW THE ARRAYS (TSAVE,ASAVE,WSAVE) ARE FULL.
C *
C *****
C
00058      DO 130 NS=1,NSEN
C
C *****
C *
C *   CALL THE SENSOR POSITION LOCATOR (SPL).
C *
C *****
C
00059      CALL SPL(NS)
00060      130 CONTINUE
C
C
C
C
C

```


NOL FORTRAN (1.0)

PROGRAM SIMULATE

```
C
C .....
C *
C * CALL OUTPUT TO PRINT RESULTS.
C *
C .....
C
```

```
00061 CALL OUTPUT
00062 NRUN=NRUN+1
00063 NRUNX=NRUNX+1
00064 IF (NRUN .GT. NUM) GO TO 101
00065 GO TO 110
00066 END
00067
```

NOL FORTRAN DIAGNOSTIC RESULTS - FOR SIMULATE

NO ERRORS

DATA VARIABLES

ALA	00001	00019	00025	00026		
ALR	00006	00048				
ALROT	00003	00035	00040			
ASAVE	00005					
AT	00001	00025	00027			
BINS	00006	00031				
BT	00001	00026	00027			
CS	00001	00030				
CT	00001	00027				
DELTA	00006	00052				
FREQ	00006					
IFLAG	00003					
INCSAVE	00005	00042				
JFLAG	00003					
KFLAG	00008	00010	00037	00039		
M	00008	00010				
MFLAG	00006	00036	00054			
NPASS	00008	00010	00012			
NREP	00005	00043				
NRUN	00006	00014	00015	00016	00017	
	00023	00029	00031	00062	00062	
NRUNX	00006	00010	00063	00063		
NSEN	00003	00050	00053	00058		
NTRG	00001	00018	00044			
NUM	00006	00064				
PI	00007	00015	00019			
PI2	00007					
PI02	00007					
RALA	00002	00019				
RALROT	00004	00040				
RANGE	00006					
RCS	00002					
RTA	00002	00024				
RTEMP	00002	00029				
RVA	00002	00023				
RX0	00004					
RXA	00002	00020				
RY0	00004					
RYA	00002	00021				
RZA	00002	00022				
SAVE	00006	00015	00016	00017	00019	
	00029	00031				
SNCPA	00006	00017				
TA	00001	00024				
TEMP	00001	00029	00030			
TLR	00006	00047				
TSAVE	00005					
VA	00001	00023				
WSAVE	00005					
X0	00003	00033				
XA	00001	00020	00027			
XK	00006	00016				

NOL FORTRAN (1.0)

PROGRAM SIMULATE

XMEAN	00006	00038		
Y0	00003	00034		
YA	00001	00021	00027	
ZA	00001	00022		

COMMON VARIABLES

NONE

PROGRAM VARIABLES

BIAS	00015	00038	00040		
COS	00025				
I	00032	00033	00034	00035	00036
	00046	00047	00048		
INPUT	00009				
J	00041	00042	00043		
NS	00050	00051	00053	00054	00058
NT	00018	00019	00019	00020	00020
	00023	00023	00024	00024	00025
	00027	00027	00027	00027	00044
OUTPUT	00061				
SENCHK	00051				
SIN	00026				
SPL	00059				
TIME	00048	00051	00052	00052	

INPUT DECK/ L.R.X

B-18

NOL FORTRAN (1.0)

```
00000          SUBROUTINE INPUT
C
C .....
C *
C *   JONATHAN VALVANO                                03/16/75 *
C *
C *   THIS ROUTINE INPUTS INITIAL PARAMETERS.
C *   ALL ANGLE INPUTS ARE IN DEGREES.
C *
C .....
00001          COMMON /TRGT// ALA(3),XA(3),YA(3),ZA(3),
00001          1  TA(3),VA(3),AT(3),BT(3),CT(3),TEMP,CS,NTRG
00002          COMMON /RTRGT// RALA(3),RXA(3),RYA(3),RZA(3),
00002          1  RTA(3),RVA(3),RTEMP,HCS
00003          COMMON /SEN// NSEN,X0(3),Y0(3),ALROT(3),IFLAG,JFLAG
00004          COMMON /HSEN// RX0(3),RY0(3),HALHOT(3)
00005          COMMON /REP// TSAVE(3,3,50),ASAVE(3,3,50),WSAVE(3,3,50),
00005          1  NREP(3,3),INCSAVE(3,3)
00006          COMMON /STATE// TLR(3),ALR(3),MFLAG(3),FREQ(2),
00006          1  BINS,XMEAN,XK,SNCPA,RANGE,DELTA,NUM,NRUN,NRUNX,SAVE(10,100)
00007          COMMON /CONST// PI,PID2,PI2
00008          COMMON /OPTIONS// KFLAG,M,NPASS
00009          DATA /CONST/ (PI=3.1415926538)
00010          DATA /CONST/ (PID2=1.570796327)
00011          DATA /CONST/ (PI2=6.283185307)
00012          DATA /REP/ (NREP(1)=9(0))
00013          DATA /STATE/ (TLR(1)=3(-50.0))
00014          DATA /STATE/ (ALR(1)=3(0.0))
C
C .....
C *
C *   READ SIMULATION OPTIONS.
C *   NUM IS THE NUMBER OF RUNS IN A SET.
C *
C .....
00015          NUM=1
00016          279  READ 280,NOYES,(SAVE(I,NUM), I=1,10)
00017          280  FORMAT (15,F5.0,F10.4,7F5.0,F10.0)
00018          NUM=NUM+1
00019          IF (NOYES.EQ. 1) GO TO 279
00021          NUM=NUM-2
C
C .....
C
C .....
C
```

NOL FORTRAN (1.0)

SUBROUTINE INPUT

```
00022      READ 200, NTRG
00023      200  FORMAT (I5)
00024      DO 225 NT=1,NTRG
C
C *****
C *
C *   READ REAL TARGET CHARACTERISTICS.
C *
C *****
C
00025      READ 220, RALAD,RXA(NT),RYA(NT),RZA(NT),RTA(NT),RVA(NT)
00026      220  FORMAT (6F10.2)
00027      RALA(NT)=RALAD*PI/180.0
00028      225  CONTINUE
00029      READ 230, NSEN
00030      230  FORMAT (I5)
00031      DO 250 NS=1,NSEN
C
C *****
C *
C *   READ REAL SENSOR CHARACTERISTICS.
C *
C *****
C
00032      READ 240, RX0(NS),RY0(NS),RALROTD
00033      240  FORMAT (3F10.2)
00034      RALROT(NS)=RALROTD*PI/180.0
00035      250  CONTINUE
00036      READ 290, RANGE,DELTA,FREQ(1),FREQ(2),RTEMP
00037      290  FORMAT (5F10.2)
C
C *****
C *
C *   NORMAL IS THE SUBROUTINE THAT GENERATES GAUSSIAN NOISE.
C *   HERE, IT IS CALLED JUST TO INITIALIZE.
C *
C *****
C
00038      CALL NORMAL (ERROR)
C
C   RCS IS THE REAL SPEED OF SOUND.
C
00039      RCS=331.36+0.6096*RTEMP
00040      RETURN
00041      END
```

NOL FORTRAN DIAGNOSTIC RESULTS - FOR INPUT

NO ERRORS

DATA VARIABLES

ALA	00001				
ALH	00006	00014			
ALROT	00003				
ASAVE	00005				
AT	00001				
BINS	00006				
BT	00001				
CS	00001				
CT	00001				
DELTA	00006	00036			
FREQ	00006	00036	00036		
IFLAG	00003				
INCSAVE	00005				
JFLAG	00003				
KFLAG	00008				
M	00008				
MFLAG	00006				
NPASS	00008				
NREP	00005	00012			
NRUN	00006				
NRUNX	00006				
NSEN	00003	00029	00031		
NTRG	00001	00022	00024		
NUM	00006	00015	00016	00018	00018
PI	00007	00009	00027	00034	
PI2	00007	00011			
PI02	00007	00010			
RALA	00002	00027			
RALROT	00004	00034			
RANGE	00006	00036			
RCS	00002	00039			
RTA	00002	00025			
RTEMP	00002	00036	00039		
RVA	00002	00025			
RX0	00004	00032			
RXA	00002	00025			
RY0	00004	00032			
RYA	00002	00025			
RZA	00002	00025			
SAVE	00006	00016			
SNCPA	00006				
TA	00001				
TEMP	00001				
TLR	00006	00013			
TSAVE	00005				
VA	00001				
WSAVE	00005				
X0	00003				
XA	00001				
XK	00006				
XMEAN	00006				
Y0	00003				

NOL FORTRAN (1.0)

SUBROUTINE INPUT

YA	00001
ZA	00001

COMMON VARIABLES

NONE

PROGRAM VARIABLES

ERROR	00038				
I	00016	00016			
NORMAL	00038				
NOYES	00016	00019			
NS	00031	00032	00032	00034	
NT	00024	00025	00025	00025	00025
RALAD	00025	00027			
RALRUTD	00032	00034			

SENCHK DECK/ L.R.X

```

00000      SUBROUTINE SENCHK (NS,TAU,NT)
      C
      C .....
      C *
      C *      JONATHAN VALVANO                      03/18/75 *
      C *
      C *      THIS ROUTINE CHECKS TO SEE IF THE SENSOR SHOULD
      C *      REPORT AT THIS TIME. IF IT DECIDES TO REPORT THEN IT MAKES
      C *      THE APPROPRIATE ENTRIES INTO THE ARRAYS. (TSAVE,ASAVE,WSAVE)
      C *
      C .....
      C
00001      COMMON /RTRGT// RALA(3),RXA(3),RYA(3),RZA(3),
00001      1  RTA(3),RVA(3),RTEMP,PCS
00002      COMMON /RSFN// RX0(3),RY0(3),RALHOT(3)
00003      COMMON /REP// TSAVE(3,3,50),ASAVE(3,3,50),WSAVE(3,3,50),
00003      1  NREP(3,3),INCSAVE(3,3)
00004      COMMON /STATE// TLR(3),ALR(3),MFLAG(3),FREQ(2),
00004      1  BINS,XMEAN,XK,SNCPA,FANGE,DELTA,NUM,NRUN,NRUNX,SAVE(10,100)
00005      COMMON /CONST// PI,PI02,PI2
      C
      C .....
      C *
      C *      THE SENSOR CAN ONLY REPORT EVERY FREQ(1) SECONDS IF THE
      C *      TARGET IS CHANGING BINS. IF IT HAS NOT BEEN AT LEAST
      C *      FREQ(1) SECONDS, CONTROL IS RETURNED TO SIMULATE,
      C *      AND THE SENSOR DOES NOT REPORT.
      C *
      C .....
00006      IF ((TAU-TLR(NS)) .LT. FREQ(1)) RETURN
      C
      C .....
      C *
      C *      TIME TAU IS WHEN THE SOUND ORIGINATES FROM THE TARGET.
      C *      RXTAU=REAL X POSITION OF THE TARGET AT TIME TAU.
      C *      RYTAU=REAL Y POSITION OF THE TARGET AT TIME TAU.
      C *      RALTAU=REAL ALPHA ANGLE TO THE TARGET AT TIME TAU.
      C *
      C .....
00008      RXTAU=RXA(NT)+RVA(NT)*(TAU-RTA(NT))*COS(PI02-RALA(NT))
00009      RYTAU=RYA(NT)+RVA(NT)*(TAU-RTA(NT))*SIN(PI02-RALA(NT))
00010      RALTAU=PI02 -ATAN((RYTAU-RY0(NS))/(RXTAU-RX0(NS)))
00011      IF (RXTAU .LT. RX0(NS)) RALTAU=RALTAU+PI
      C
      C .....
      C *
      C *      DXYZ IS THE DISTANCE FROM THE SENSOR TO THE TARGET. IT IS
      C *      THIS DISTANCE THAT THE SOUND TRAVELS (FROM TARGET TO SENSOR)
      C *
      C .....
00013      DXYZ=SQRT((RXTAU-RX0(NS))**2+(RYTAU-RY0(NS))**2+RZA(NT)**2)

```

```

C
C .....
C *
C *   IF THE TARGET IS OUT OF RANGE FOR THIS SENSOR,
C *   SET MFLAG TO 1 AND RETURN.   NO REPORT.
C *
C .....
C
00014       IF (WXYZ .LT. RANGE) GO TO 300
00016       MFLAG(NS)=1
00017       RETURN
00018   300   MFLAG(NS)=0
C
C .....
C *
C *   TIME IS THE TIME WHEN THE SOUND REACHS THE SENSOR.
C *   IT IS THE REPORT TIME.
C *
C .....
C
00019       TIME=TAU+DXYZ/RCS
C
C .....
C *
C *   ALTAU WILL BE THE REPORT BEARING
C *
C .....
C
00020       ALTAU=RALTAU+RALROT(NS)
C
C .....
C *
C *   CALCULATE SIGNAL/NOISE (SN)
C *
C *   ADD GAUSSIAN ERROR.
C *
C .....
C
00021       SN=SNCPA/(WXYZ+1.0)
00022       XSIGMA=XK/SQRT(SN)
00023       CALL NORMAL (ERROR)
00024       ALTAU=ALTAU+XMEAN+XSIGMA*ERROR
C
C .....
C *
C *   ALTAU IS ADJUSTED TO FIT INTO A BEARING BIN.
C *
C .....
C
00025       ALTAU=(PI2/BINS)*FLOAT (IFIX (BINS*ALTAU/PI2+0.5))
C
C
C

```

NOL FORTRAN (1.0)

SUBROUTINE SENCHK (NS,TAU,NT)

```
C
C *****
C *
C *   W IS THE WEIGHTING FACTOR PASSED FROM THE SENSOR.
C *
C *****
00026      W=.001
C
C *****
C *
C *   IF THE BEARING HAS NOT CHANGED SINCE THE LAST REPORT
C *   (ALTAU=ALR), THEN THE SENSOR MUST WAIT AT LEAST FREQ(2)
C *   SINCE THE LAST REPORT. WE KNOW THE SENSOR HAS WAITED
C *   FREQ(1) SECONDS (FROM ABOVE), SO WE WILL REPORT IF EITHER
C *   1. THE BEARING HAS CHANGED
C *   OR  2. THE TIME DIFFERENCE .GT. FREQ(2)
C *
C *****
00027      IF (( TAU-TLR(NS)) .LT. FREQ(2) .AND.
00027      1  ALTAU .EQ. ALR(NS)) RETURN
C
C *****
C *
C *   THE SENSOR IS NOW REPORTING.
C *
C *****
00029      IF (NREP(NS,NT) .GE. 50) RETURN
00031      TLR(NS)=TAU
00032      ALR(NS)=ALTAU
00033      NREP(NS,NT)=NREP(NS,NT)+1
00034      N=NREP(NS,NT)
00035      TSAVE(NS,NT,N)=TIME
00036      ASAVE(NS,NT,N)=ALTAU
00037      WSAVE(NS,NT,N)=W
00038      RETURN
00039      END
```

NOL FORTRAN DIAGNOSTIC RESULTS - FOR SENCHK

NO ERRORS

DATA VARIABLES

ALR	00004	00027	00032		
ASAVE	00003	00036			
BINS	00004	00025	00025		
DELTA	00004				
FREQ	00004	00006	00027		
INCSAVE	00003				
MFLAG	00004	00016	00018		
NREP	00003	00029	00033	00033	00034
NRUN	00004				
NRUNX	00004				
NUM	00004				
PI	00005	00012			
PI2	00005	00025	00025		
PID2	00005	00008	00009	00010	
RALA	00001	00008	00009		
RALROT	00002	00020			
RANGE	00004	00014			
RCS	00001	00019			
RTA	00001	00008	00009		
RTEMP	00001				
RVA	00001	00008	00009		
RX0	00002	00010	00011	00013	
RXA	00001	00008			
RY0	00002	00010	00013		
RYA	00001	00009			
RZA	00001	00013			
SAVE	00004				
SNCPA	00004	00021			
TLR	00004	00006	00027	00031	
TSAVE	00003	00035			
WSAVE	00003	00037			
XK	00004	00022			
XMEAN	00004	00024			

COMMON VARIABLES

NONE

PROGRAM VARIABLES

ALTAU	00020	00024	00024	00025	00025
ATAN	00010				
COS	00008				
DXYZ	00013	00014	00019	00021	
ERROR	00023	00024			
FLOAT	00025				
IFIX	00025				
N	00034	00035	00036	00037	
NORMAL	00023				
NS	00000	00006	00010	00010	00011

NOL FORTRAN (1.0)

SUBROUTINE SENCHK (NS,TAU,NT)

	00020	00027	00027	00029	00031
	00035	00036	00037		
NT	00000	00008	00008	00008	00008
	00013	00029	00033	00033	00034
RALTAU	00010	00012	00012	00020	
RXTAU	00008	00010	00011	00013	
RYTAU	00009	00010	00013		
SIN	00009				
SN	00021	00022			
SWRT	00013	00022			
TAU	00000	00006	00008	00009	00019
TIME	00019	00035			
W	00026	00037			
XSIGMA	00022	00024			

OUTPUT DECK/ L.R.X

NOL FORTRAN (1.0)

SUBROUTINE OUTPUT

```

C
00030      DO 610 NT=1,NTRG
00031      PRINT 605, NT
00032      505      FORMAT (1H3,10X,5MTARG ,15,
00032      1      56H      XA      YA      ZA      VA COURSE      TEMP      CS)
00033      RALAU=RALA(NT)*180.0/PI
00034      PRINT 606, RXA(NT),RYA(NT),RZA(NT),RVA(NT),RALAU,RTEMP,RCS
00035      606      FORMAT (1H ,10X,10HREAL      ,7F8.2)
00036      ALAU=ALA(NT)*180.0/PI
00037      PRINT 607, XA(NT),YA(NT),ZA(NT),VA(NT),ALAU,TEMP,CS
00038      607      FORMAT (1H ,10X,10HINPUTED      ,7F8.2)
00039      610      CONTINUE
00040      DO 650 NS=1,NSEN
00041      PRINT 620, NS
00042      620      FORMAT (1H3,10X,14HSENSOR NUMBER ,15)
00043      PRINT 621
00044      621      FORMAT (1H ,10X,20X,
00044      1      36H      X0      Y0      ALROT)
00045      RALROT=RALROT(NS)*180.0/PI
00046      PRINT 622, RX0(NS),RY0(NS),RALROT
00047      622      FORMAT (1H ,10X,4HREAL,16X,3F12.5)
00048      ALROT=ALROT(NS)*180.0/PI
00049      PRINT 623, X0(NS),Y0(NS),ALROT
00050      623      FORMAT (1H ,10X,10HCALCULATED,10X,3F12.5)
00051      DX=RX0(NS)-X0(NS)
00052      DY=RY0(NS)-Y0(NS)
00053      DAL=RALROT-ALROT
00054      PRINT 624, DX,DY,DAL
00055      624      FORMAT (1H ,10X,10HDIFFERENCE,10X,3F12.5)
00056      DR=SQRT(DX**2+DY**2)
00057      PRINT 625, DR
00058      625      FORMAT (1H,10X,12HRADIAL ERROR,8X,F12.5)
00059      PRINT 616, (INCSAVE(NS,I), I=1,3)
00060      616      FORMAT (1H ,10X,10HINCREMENTS,10X,3I12)
00061      650      CONTINUE
00062      RETURN
00063      END
```

NOL FORTRAN DIAGNOSTIC RESULTS - FOR OUTPUT

NO ERRORS

B-31

COPY AVAILABLE TO DDC DOES NOT
PERMIT FULLY LEGIBLE PRODUCTION

DATA VARIABLES

ALA	00001	00036			
ALR	00006				
ALROT	00003	00048			
ASAVE	00005				
AT	00001				
BINS	00006	00028			
BT	00001				
CS	00001	00037			
CT	00001				
DELTA	00006	00028			
FREQ	00006	00026	00026		
IFLAG	00003				
INCSAVE	00005	00059			
JFLAG	00003				
KFLAG	00007	00011	00014		
M	00007	00019			
MFLAG	00006				
NPASS	00007	00019			
NREP	00005				
NRUN	00006				
NRUNX	00006	00009			
NSEN	00003	00040			
NTRG	00001	00030			
NUM	00006				
PI	00008	00021	00033	00036	00045
PI2	00008				
PID2	00008				
RALA	00002	00033			
RALROT	00004	00045			
RANGE	00006	00022			
RCS	00002	00034			
RTA	00002				
RTEMP	00002	00034			
RVA	00002	00034			
RX0	00004	00046	00051		
RXA	00002	00034			
RY0	00004	00046	00052		
RYA	00002	00034			
RZA	00002	00034			
SAVE	00006				
SNCPA	00006	00024			
TA	00001				
TEMP	00001	00037			
TLR	00006				
TSAVE	00005				
VA	00001	00037			
WSAVE	00005				
X0	00003	00049	00051		
XA	00001	00037			
XK	00006	00024			
XMEAN	00006	00021			
Y0	00003	00049	00052		

NOL FORTRAN (1.0)

SUBROUTINE OUTPUT

YA	00001	00037
ZA	00001	00037

COMMON VARIABLES

NONE

PROGRAM VARIABLES

ALAD	00036	00037		
ALROTD	00048	00049	00053	
DAL	00053	00054		
DR	00056	00057		
DX	00051	00054	00056	
DY	00052	00054	00056	
I	00059	00059		
NS	00040	00041	00045	00046 00046
	00051	00052	00052	00059
NT	00030	00031	00033	00034 00034
	00037	00037	00037	
RALAD	00033	00034		
RALROTD	00045	00046	00053	
SWRT	00056	.		
XMEAND	00021	00022		

B-33

COPY AVAILABLE TO DDC DOES NOT
PERMIT FULLY LEGIBLE PRODUCTION

SPL DECK/ L,R,X

B-34

COPY AVAILABLE TO DDC DOES NOT
PERMIT FULLY LEGIBLE PRODUCTION


```

C *****
C *
C *   WHEN CALCULATING THE BIAS IT IS BETTER NOT TO USE
C *   SEQUENTIAL REPORTS BECAUSE THE ANGLE DIFFERENCES ARE
C *   EITHER TOO SMALL OR ZERO. THEREFORE WE WILL INCREMENT
C *   THROUGH THE ARRAYS STEPPING MORE THAN ONE. INC IS THE
C *   VALUE OF THE INCREMENT FOR EACH TARGET THE SENSOR SA#.
C *   IF INC(1)=4 THEN WHEN WE LOOK AT THE REPORTS OF TARGET 1,
C *   WE WILL STEP THROUGH BY 4 (1,5,9,13,...).
C *   IN ORDER TO USE AS MUCH OF THE DATA AS POSSIBLE, WE
C *   WILL STEP THROUGH THE ARRAYS AGAIN BUT STARTING WITH 2,
C *   THEN 3, THEN 4. IF NREP(NS,1)=50 THEN INC(1) WOULD BE 13
C *   AND WE WOULD LOOK AT THE REPORTS IN THIS ORDER--
C *   1,14,27,40 2,15,28,41 3,16,29,41 .... 13,26,39
C *   BIAS CALCULATIONS USE REPORTS IN GROUPS OF THREE.
C *   POSITION CALCULATION S USE EACH REPORT SEPERATELY.
C *****
C
00007      DO 400 NT=1,NTRG
00008          INCSAVE(NS,NT)=1+NREP(NS,NT)/M
00009          CONTINUE
C
C *****
C *
C *   THE POSITION CALCULATION CAN NOT BE DONE WITHOUT A BIAS
C *   CALCULATION . BUT THE BIAS CALCULATION CAN BE DONE WITH
C *   OR WITHOUT A POSITION CALCULATION . THE BIAS CALCULATION
C *   IS MORE ACCURATE WHEN POSITION IS KNOWN. TO KEEP
C *   TRACK OF WHICH CALCULATIONS HAVE BEEN DONE FLAGS
C *   ARE USED.
C *
C *           IFLAG=0   MEANS NO BIAS CALCULATION
C *           IFLAG=1   MEANS A BIAS CALCULATION
C *           JFLAG=0   MEANS NO POSITION CALCULATION
C *           JFLAG=1   MEANS A POSITION CALCULATION
C *           KFLAG=0   MEANS BIAS TO BE CALCULATED
C *           KFLAG=1   MEANS BIAS HAS BEEN INPUTED
C *****
00010      401      IFLAG=0
00011              JFLAG=0
00012              DO 490 NP=1,NPASS
C *****
C *
C *   EACH PASS STARTS WITH FRESH SUMS BUT USES THE CALCULATIONS
C *   FROM THE PREVIOUS PASS.
C *****
00013          A12=0.0
00014          B12=0.0
00015          A1B1=0.0
00016          A1C1=0.0
00017          B1C1=0.0
00018          BSI=0.0
00019          WI=0.0
    
```

COPY AVAILABLE TO DDC DOES NOT PERMIT FULLY LEGIBLE PRODUCTION


```

C
C
C .....
C *
C * THE BIAS ALGORITHM WILL NOT WORK IF THETA1 EQUALS ZERO.
C * THE INCREMENT IS INCREASED AND WE START OVER AGAIN.
C *
C .....
C
C
00036 IF (THETA1 .NE. 0.0) GO TO 410
00038 INCSAVE(NS,NT)=INCSAVE(NS,NT)+1
00039 GO TO 401
00040 410 DO 460 N3=NSTART,NRPT,N
00041 CALL ENTER (NS,NT,N3,T3,X3,Y3,ALPHA3,N3)
00042 IF (KFLAG .EQ. 1) GO TO 460
00044 D23=DSQRT((X3-X2)**2+(Y3-Y2)**2)
00045 THETA2=ALPHA3-ALPHA2
C
C .....
C *
C * SHIFT WILL SHIFT THETA2 BETWEEN -PI AND +PI
C *
C .....
00046 CALL SHIFT (THETA2,0.0)
C
C .....
C *
C * THE ALGORITHM WILL NOT WORK IF
C * THETA1 EQUALS ZERO
C * THETA2 EQUALS ZERO
C * THETA1 AND THETA2 HAVE OPPOSITE SIGNS.
C * IF ANY OF THESE FAULTS OCCUR THE ALGORITHM IS SKIPPED.
C *
C .....
00047 IF (THETA1+THETA2 .LE. 0.0) GO TO 460
C
C
C
C

```

COPY AVAILABLE TO DDC DOES NOT PERMIT FULLY LEGIBLE PRODUCTION


```

C
C .....
C *
C * WE HAVE NOW FINISHED ONE LOOP THROUGH THE ARRAYS.
C * CONTROL WILL BE HERE ABOUT INC(NT) TIMES FOR EACH
C * TARGET AND FOR EACH PASS AND FOR EACH SENSOR.
C *
C .....
C .....
C *
C * IF WI EQUALS ZERO THAT MEANS NO BIAS CALCULATIONS HAVE
C * BEEN DONE. IF THIS IS SO THEN NO AVERAGING CAN BE DONE.
C *
C .....
00068 C IF (WI .EQ. 0.0 .OR. KFLAG .EQ. 1) GO TO 465
C .....
C *
C * CALCULATE THE AVERAGE BIAS AND SET IFLAG.
C *
C .....
00070 C ALROT(NS)=PSI/WI
00071 C IFLAG=1
00072 C 465 DENOM=A12**2-A1B1**2
C .....
C *
C * IF DENOM EQUALS ZERO THEN WE HAVE CALCULATED ONLY ONE OR
C * NO LINES THROUGH THE SENSOR. IN EITHER CASE A SENSOR
C * POSITION CALCULATION CAN NOT BE DONE AND IS SKIPPED.
C *
C .....
00073 C IF (DENOM .EQ. 0.0) GO TO 470
C .....
C *
C * THE SENSOR POSITION LOCATION IS CALCULATED. AND JFLAG SET.
C *
C .....
00075 C X0(NS)=(B1C1*A1B1-A1C1*B1I2)/DENOM
00076 C Y0(NS)=(A1C1*A1B1-B1C1*A1I2)/DENOM
00077 C JFLAG=1
00078 C 470 CONTINUE
00079 C 480 CONTINUE
00080 C 490 CONTINUE
00081 C RETURN
00082 C END

```

NOL FORTRAN DIAGNOSTIC RESULTS - FOR SPL

B-41

NO ERRORS

COPY AVAILABLE TO DDC DOES NOT PERMIT FULLY LEGIBLE PRODUCTION

DATA VARIABLES

AI2	00005	00013	00072	00076	
AIBI	00005	00015	00072	00075	00075
AICI	00005	00016	00075	00076	
ALA	00001	00054			
ALROT	00002	00055	00070		
ASAVE	00003				
AT	00001				
BI2	00005	00014	00072	00075	
BICI	00005	00017	00075	00076	
BT	00001				
CS	00001				
CT	00001				
IFLAG	00002	00010	00071		
INCSAVE	00003	00008	00022	00038	00038
JFLAG	00002	00011	00077		
KFLAG	00005	00031	00042	00064	
M	00006	00008			
NPASS	00005	00012			
NREP	00003	00008	00021		
NSEN	00002				
NTRG	00001	00007	00020		
PI	00004	00053	00054		
PI2	00004				
PID2	00004				
TA	00001				
TEMP	00001				
TSAVE	00003				
VA	00001				
WSAVE	00003				
X0	00002	00075			
XA	00001				
Y0	00002	00076			
YA	00001				
ZA	00001				

COMMON VARIABLES

NONE

PROGRAM VARIABLES

ALPHA1	00029	00034	00054	00063	
ALPHA2	00030	00034	00045	00063	00064
ALPHA3	00041	00045	00064		
ATAN	00051				
BS	00054	00055	00056		
BSI	00018	00056	00056	00070	
COS	00050				
D12	00033	00050	00065		
D23	00044	00050	00065		

NOL FORTRAN (1.0)

SUBROUTINE SPL(NS)

DENOM	00050	00051	00052	00072	00073
ENTER	00029	00030	00041		
N	00022	00023	00029	00027	00028
N1	00025	00027	00029	00058	00059
N2	00027	00028	00030		
N3	00040	00041	00059		
NFIRST	00025	00026			
NP	00012				
NRPT	00021	00023	00040		
NS	00000	00008	00008	00021	00022
	00041	00055	00070	00075	00076
NSTART	00028	00040			
NT	00007	00008	00008	00020	00021
	00038	00041	00054		
PHI	00051	00053	00053	00054	
SHIFT	00035	00046	00055		
SIN	00050	00050	00051		
SQRT	00033	00044			
T1	00029				
T2	00030				
T3	00041				
THETA1	00034	00035	00038	00047	00049
THETA12	00049	00050	00051		
THETA2	00045	00046	00047	00049	00050
w1	00029				
w2	00030	00056	00057	00058	
w3	00041	00056			
wI	00019	00057	00057	00058	00070
x1	00029	00033			
x2	00030	00033	00044	00050	
x3	00041	00044	00060		
Y1	00029	00033			
Y2	00030	00033	00044	00051	
Y3	00041	00044	00061		

AD-A032 850 NAVAL SURFACE WEAPONS CENTER WHITE OAK LAB SILVER SP--ETC F/G 17/1
SENSOR POSITION LOCATOR.(U)

UNCLASSIFIED

2 OF 2
AD
A032 850

NSWC/WOL/TR 76-8

NL



ENTER DECK/ L.R.X

B-44

COPY AVAILABLE TO DDC DOES NOT
PERMIT FULLY LEGIBLE PRODUCTION


```

00007      T=TSAVE(NS,NT,NR)
00008      ALPHA=ASAVE(NS,NT,NR)
00009      W=WSAVE(NS,NT,NR)
00010      CALL TXY (XT,YT,T,NT)

C
C *****
C *
C *   IF WE ARE GOING TO CALCULATE THE BIAS (KFLAG=0) AND
C *   IF A BIAS CALCULATION HAS NOT BEEN DONE THEN RETURN
C *
C *****
C
00011      IF (IFLAG .EQ. 0 .AND. KFLAG .EQ. 0) RETURN

C
C *****
C *
C *   THE RATIO IS USED IN THE ALGORITHM TO ACCOUNT FOR THE
C *   SPEED OF SOUND. THE TARGET SPEED (VA) MUST BE LESS THAN
C *   THE SPEED OF SOUND (CS). THIS ALONE CORRECTS ONLY THE
C *   TWO DIMENSIONAL CASE. IF A SENSOR CALCULATION HAS BEEN
C *   DONE BEFORE (JFLAG=1) THEN THIS ALGORITHM CAN CORRECT
C *   THE THREE DIMENSIONAL CASE. IF NO SENSOR POSITION
C *   LOCATION HAS BEEN DONE (JFLAG=0) THEN JUST USE THE P-D.
C *
C *****
C
00013      BETA=PI+ALA(NT)-(ALPHA-ALROT(NS))
00014      RATIO=CS/VA(NT)
00015      RXYTAU=10000.0
00016      IF (JFLAG .EQ. 0) GO TO 500

C
C *****
C *
C *   (XP,YP) IS THE POINT ON THE TARGET TRACK AT CPA
C *   THE POINT IS USED IN THE ALGORITHM TO CORRECT FOR THE
C *   SPEED OF SOUND DUE TO THE ALTITUDE. THE RATIO IS
C *   RECALCULATED TAKING THE ALTITUDE INTO EFFECT.
C *   RXYTAU IS THE TWO DIMENSIONAL RADIAL DISTANCE FROM
C *   THE SENSOR TO THE TARGET AT TIME TAU (TIME WHEN THE SOUND
C *   ORIGINATED.) RXYTAU IS USED TO CONSTRUCT A W.
C *
C *****
C
00018      D=AT(NT)*Y0(NS)-BT(NT)*X0(NS)
00019      DENOM=AT(NT)**2+BT(NT)**2
00020      XP=-(AT(NT)*CT(NT)+BT(NT)*D)/DENOM
00021      YP=(AT(NT)*D-BT(NT)*CT(NT))/DENOM
00022      RXYTAU=(SQRT((X0(NS)-XP)**2+(Y0(NS)-YP)**2))/
00022      1 ABS(SIN(BETA))
00023      RATIO=RATIO*RXYTAU/SQRT(ZA(NT)**2+RXYTAU**2)
00024      IF (RXYTAU .GT. 5000.0) RXYTAU=5000.0
00026      W=1000.0*W*EXP(-(RXYTAU/1000.0)**2)

C
C

```

COPY AVAILABLE TO DDC DOES NOT PERMIT FULLY LEGIBLE PRODUCTION

NOL FORTRAN (1.0)

SUBROUTINE ENTER (NS,NT,NR,T,XT,YT,ALPHA,.)

```

C .....
C *
C * THETA IS THE CORRECTION ANGLE DUE TO THE SPEED OF SOUND.
C * CALPHA IS THE CORRECTED REPORT BEARING.
C * ALPHA IS CORRECTED AND PASSED BACK TO THE BIAS ALGORITHM
C *
C * CALPHA IS USED TO CALCULATE A,B,C OF THE LINE A*X+B*Y+C=0
C * THIS LINE GOES THROUGH THE SENSOR AND THE TARGET
C * THE ALGORITHM WILL EVENTUALLY (INSPL) FIND THE POINT
C * (X0,Y0) WHICH MINIMIZES THE DISTANCE FROM THAT POINT
C * TO EACH LINE. A WEIGHTING FACTOR IS USED IN THE LEAST
C * SQUARES ALGORITHM.
C .....
C
00027 500 THETA=ATAN(SIN(THETA)/(RATIO-COS(THETA)))
00028 CALPHA=(ALPHA-ALROT(NS))-THETA
00029 ALPHA=ALPHA-THETA
C .....
C *
C * CALCULATE A,B,C TO FORM THE LINE A*X+B*Y+C=0
C *
C .....
C
00030 A=COS(CALPHA)
00031 B=-SIN(CALPHA)
00032 C=-(A*XT+B*YT)
C .....
C *
C * MAKE THE SUMS.
C *
C .....
C
00033 A2PB2=(A**2+B**2)*RX*YTAU**2
00034 AI2=AI2+W*A**2/A2PB2
00035 BI2=BI2+W*B**2/A2PB2
00036 AIBI=AIBI+W*A*B/A2PB2
00037 AICI=AICI+W*A*C/A2PB2
00038 BICI=BICI+W*B*C/A2PB2
00039 RETURN
00040 END

```

NOL FORTRAN DIAGNOSTIC RESULTS - FOR ENTER

NO ERRORS

DATA VARIABLES

AI2	00005	00034	00034		
AIB1	00005	00036	00036		
AICI	00005	00037	00037		
ALA	00001	00013			
ALROT	00002	00013	00028		
ASAVE	00003	00008			
AT	00001	00018	00019	00020	00021
B12	00005	00035	00035		
BICI	00005	00038	00038		
BT	00001	00018	00019	00020	00021
CS	00001	00014			
CT	00001	00020	00021		
IFLAG	00002	00011			
INCSAVE	00003				
JFLAG	00002	00016			
KFLAG	00006	00011			
M	00006				
NPASS	00006				
NREP	00003				
NSEN	00002				
NTRG	00001				
PI	00004	00013			
P12	00004				
PIU2	00004				
TA	00001				
TEMP	00001				
TSAVE	00003	00007			
VA	00001	00014			
WSAVE	00003	00009			
X0	00002	00018	00022		
XA	00001				
Y0	00002	00018	00022		
YA	00001				
ZA	00001	00023			

COMMON VARIABLES

NONE

PROGRAM VARIABLES

A	00030	00032	00033	00034	00036
AZPB2	00033	00034	00035	00036	00037
AHS	00022				
ALPHA	00000	00008	00013	00028	00029
ATAN	00027				
B	00031	00032	00033	00035	00036
HETA	00013	00022	00027	00027	
C	00032	00037	00038		
CALPHA	00028	00030	00031		

NOL FORTRAN (1.0)

SUBROUTINE ENTER (NS,NT,NR,T,XT,YT,ALPHA,W)

COS	00027	00030			
D	00018	00020	00021		
DENOM	00019	00020	00021		
EXP	00026				
NR	00000	00007	00008	00009	
NS	00000	00007	00008	00009	00013
	00028				
NT	00000	00007	00008	00009	00010
	00019	00019	00020	00020	00020
RATIO	00014	00023	00023	00027	
RXYTAU	00015	00022	00023	00023	00024
SIN	00022	00027	00031		
SQRT	00022	00023			
T	00000	00007	00010		
THETA	00027	00028	00029		
TXY	00010				
W	00000	00009	00020	00026	00034
XP	00020	00022			
XT	00000	00010	00032		
YP	00021	00022			
YT	00000	00010	00032		

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SHIFT DECK/ L.R.X

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10L FORTRAN (1.0)

00000

SUBROUTINE SHIFT (ANGLE,BIAS)

```
C .....  
C *  
C * JONATHAN VALVANO 03/16/75 *  
C *  
C * THIS ROUTINE SHIFTS THE ANGLE BETWEEN *  
C * (BIAS-PI) TO (BIAS+PI) *  
C *  
C .....  
C
```

00001

COMMON /CONST// PI,PI02,PI2

00002

800 IF (BIAS-PI .LT. ANGLE) GO TO 810

00004

ANGLE=ANGLE+PI2

00005

GO TO 800

00006

810 IF (BIAS+PI .GT. ANGLE) RETURN

00008

ANGLE=ANGLE-PI2

00009

GO TO 810

00010

END

NOL FORTRAN DIAGNOSTIC RESULTS - FOR SHIFT

10 ERRORS

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NOL FORTRAN (1.0)

SUBROUTINE SHIF1 (ANGLE,BIAS)

DATA VARIABLES

PI	00001	00002	00006
PI2	00001	00004	00006
PI02	00001		

COMMON VARIABLES

NONE

PROGRAM VARIABLES

ANGLE	00000	00002	00004	00004	00006
BIA5	00000	00002	00006		

TXY

DECK/

L.R.A

B-53

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NOL FORTRAN (1.0)

SUBROUTINE TXY (X,Y,T,NT)

DATA VARIABLES

ALA	00001	00003
AT	00001	
BT	00001	
CS	00001	
CT	00001	
NTRG	00001	
PI	00002	
PI2	00002	
PID2	00002	00003
TA	00001	00004
TEMP	00001	
VA	00001	00004
XA	00001	00005
YA	00001	00006
ZA	00001	

COMMON VARIABLES

NONE

PROGRAM VARIABLES

ANGLE	00003	00005	00006		
COS	00005				
DIST	00004	00005	00006		
NT	00000	00003	00004	00004	00005
SIN	00005				
T	00000	00004			
X	00000	00005			
Y	00000	00006			

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TXY
NORMAL

DECK/
DECK/

L.R.X
L.R.X

000

B-56

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L FORTRAN (1.0)

00000 SUBROUTINE NORMAL(X)

```
C
C FORTRAN CALL LINE:
C
C CALL NORMAL(X)
C
C THIS PROGRAM USES THE POLAR METHOD TO GENERATE NORMALLY DISTRIBUTED
C RANDOM NUMBERS OF ZERO MEAN AND UNIT VARIANCE. THE NUMBERS ARE
C GENERATED TWO AT A TIME, SO EVERY OTHER CALL REQUIRES NO WORK.
C THE LOCAL INTEGER MODE VARIABLE M WILL NORMALLY NOT BE ONE AT LOAD
C TIME. IF THIS IS NOT THE CASE FOR A PARTICULAR FORTRAN SYSTEM, THE
C USER CAN INITIALIZE THE PROGRAM VARIABLES BY MAKING AN INITIAL CALL
C AND DISREGARDING THE RESULT.
C
C THE SUBROUTINE UNIFORM USED IN THE PROGRAM SHOULD GENERATE RANDOM
C NUMBERS UNIFORMLY DISTRIBUTED BETWEEN ZERO AND ONE.
C
```

```
00001 IF(M .EQ. 1) GO TO 20
00002 M=1
00003 10 CALL UNIFORM(Y)
00004 U1=2.*Y-1.
00005 CALL UNIFORM(Y)
00006 U2=2.*Y-1.
00007 S=U1*U1+U2*U2
00008 IF(S .GE. 1.) GO TO 10
00009 S=SQRT(-2.*ALOG(S)/S)
00010 X=U1*S
00011 RETURN
00012 20 M=0
00013 X=U2*S
00014 RETURN
00015 END
00016
00017
```

NOL FORTRAN DIAGNOSTIC RESULTS - FOR NORMAL

ERRORS

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NOL FORTRAN (1.0)

SUBROUTINE NORMAL (X)

DATA VARIABLES

NONE

COMMON VARIABLES

NONE

PROGRAM VARIABLES

ALOG	00011				
M	00001	00003	00014		
S	00009	00009	00011	00011	00011
SURTF	00011				
U1	00009	00002	00009	00012	
U2	00007	00006	00003	00013	
UNIFORM	00004	00006			
X	00010	00012	00015		
Y	00004	00005	00006	00007	

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SCMP

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NOL COMPASS (1.0)
UNIFORM DECK/ L.X

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NOL COMPASS (1.0)

UNIFORM

ENTRY POINT SYMBOLS
UNIFORM 00000

PROGRAM 00052
COMMON
DATA

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ENTRY UNIFORM

FORTRAN CALL LINE:

CALL UNIFORM(X)

IF X IS GREATER THAN OR EQUAL TO ZERO WHEN THE CALL IS EXECUTED, X IS REPLACED WITH A RANDOM NUMBER UNIFORMLY DISTRIBUTED BETWEEN 0 AND 1. THE COMPUTATION USES 36 BIT INTEGER ARITHMETIC, AND CONVERTS TO FLOATING POINT. THE ALGORITHM IS THE LINEAR CONGRUENTIAL METHOD WITH A MODULUS OF 2^{36} , A MULTIPLIER OF 5^{15} , AND AN INCREMENT OF 7261067005. THESE PARAMETERS HAVE BEEN EXTENSIVELY TESTED, AND YIELD A MAXIMUM LENGTH SEQUENCE WITH GOOD SPECTRAL PROPERTIES.

IF X IS LESS THAN ZERO WHEN THE CALL IS EXECUTED, -X IS CONVERTED TO A 36 BIT FIXED POINT FRACTION, AND THE MOST SIGNIFICANT 36 BITS ARE USED TO INITIALIZE THE RANDOM NUMBER GENERATOR. IF -X IS GREATER THAN OR EQUAL TO 1 OR LESS THAN 2^{-36} , THE SEQUENCE WILL START WITH ZERO. THE STARTING VALUE IS CONVERTED TO FLOATING POINT AND RETURNED IN X, SO THAT X IS ALWAYS A FRACTION BETWEEN 0 AND 1 ON RETURN.

ENTER, GET X, AND BRANCH TO INITIALIZATION IF NEGATIVE.

```

77777 0 UNIFORM UJP      **      ENTRY POINT
P00000 1          LDI      UNIFORM,1 PARAMETER ADDRESS TO B1
00000 1          LOAQ,I   0,1      X TO AQ
P00024 3          AZJ,LT   INIT     INITIALIZE IF X LT ZERO

```

IF X GE ZERO ON CALL, GENERATE A NEW RANDOM NUMBER.

```

P00042 0          LDAQ      Y      NEWY=(A*OLOY+2*C)MOD2**36
P00044 0          MUAQ      A
00000 3          EAQ
07777 2          JNA      7777B
P00046 0          ADAQ      C2
07777 2          ANA      7777B
P00042 0 CONT     STAQ      Y      SAVE Y FOR NEXT TIME

```

CONVERT THE 36 BIT NUMBER IN AQ TO A FLOATING POINT FRACTION.

```

02013 2          SCAQ      2013B,2 SCALE AQ,BIASED EXPONENT TO B2
P00022 0          AZJ,EQ   RETURN  ZERO IS A SPECIAL CASE
02000 2          ISG      2000B,2 TEST FOR NON-NEGATIVE EXPONENT
77776 2          INI      -1,2    ADJUST BIAS IF NEGATIVE
00031 0          SHAQ      25     MERGE FRACTION AND EXPONENT
40000 2          AIA      2
00014 0          SHAQ      12     AQ IS NOW A FLOATING POINT FRACTION

```

STORE AQ IN X, AND RETURN TO CALLING PROGRAM.

```

00000 1 RETURN   STAQ,I   0,1     STORE AQ IN X

```

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```

00001 1      UJP      1.1      RETURN TO PARAMETER ADDRESS + 1
      IF X LT ZERO ON CALL. CONVERT -X TO FIXED POINT AND USE IT TO
      INITIALIZE THE RANDOM NUMBER GENERATOR.

77777 0  INIT      XOA,S      77777B      NEGATE AQ
77777 1      YOG,S      77777B
00044 0      SHAQ      30          BIASFD EXPONENT TO 92
00000 2      TAI        2
03777 2      ANI        37770.2
00014 0      SHAQ      12          FRACTION TO AQ
07777 2      ANA        7777B
02001 2      ISG        20010.2   SKIP IF EXPONENT GE 1
01735 2      ISG        17350.2   SKIP IF EXPONENT GE -34
P00051 0     LDAQ      ZERO        FORCE ZERO IF EXPONENT OUT OF ROUNDS
01777 2      ISG        17770.2   SKIP IF EXPONENT NON-NEGATIVE
76000 2      SHAQ      -17770.2   EXPONENT IS SHIFT COUNT. IF NEGATIVE
77776 1      ANQ,S     77770B     CLEAR LSF TO RETAIN 30 BITS
P00012 0     UJP      CUNT        USE AQ FOR NEW Y IN MAIN LOOP

```

CONSTANTS AND VARIABLES

```

100      Y      OCT      0.0
100
132      A      OCT      3432.77244615 THIS IS 5**15
115
141      C2     OCT      1541.45427232 THIS IS 2*7261067005
132
100      ZERO   OCT      0.0
100

```

END

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