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AN ENVIRONMENTAL HEAT TRANSFER STUDY OF
A ROCKET MOTOR STORAGE CONTAINER SYSTEM

Allen H. Wirzburge:

Naval Postgraduate School
Monterey, California

December 1972

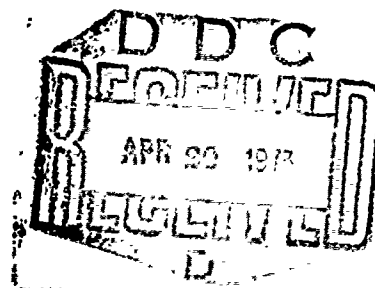
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THESIS

AN ENVIRONMENTAL HEAT TRANSFER STUDY
OF
A ROCKET MOTOR STORAGE CONTAINER SYSTEM

by

Allen Henry Wirzburger

Thesis Advisor:

T. E. Cooper

December 1972

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| 13. ABSTRACT | | | |
| <p>The heat transfer characteristics of a rocket motor storage container system have been investigated using analytical and experimental techniques. Analytically, both closed form and numerical solutions have been developed. These solutions may be used to determine maximum temperatures and temperature gradients within the rocket motor. Comparison between theoretical and experimental values of temperature are within the estimated experimental uncertainties of $\pm 3^\circ\text{F}$. It is proposed that the theoretical solutions can be used to thermally optimize container design.</p> <p>A secondary investigation was carried out to determine the feasibility of using cholesteric liquid crystals, a temperature sensitive material, to thermally map the surface of the container. The crystals appear to remain stable under desert type conditions and produce brilliantly colored displays of the temperature field.</p> | | | |

14.

KEY WORDS

LINK A

LINK B

LINK C

ROLE

WT

ROLE

WT

ROLE

WT

heat transfer
 conduction
 radiation
 convection
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 TRUMP
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An Environmental Heat Transfer Study
of
A Rocket Motor Storage Container System

by

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Submitted in partial fulfillment of the
requirements for the degree of

MASTER OF SCIENCE IN MECHANICAL ENGINEERING

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TABLE OF SYMBOLS

$a = \sqrt{\frac{\omega r_o^2}{\alpha}}$ = conduction parameter
 A_n = area of surface n $\frac{\text{sq in}}{\text{in}}$
 $B = \frac{1}{T}$ = volume coefficient of expansion $\frac{1}{^\circ R}$
 $c =$ specific heat $\frac{\text{BTU}}{\text{lbm } ^\circ F}$
 $D_n = D_n' + D_n'' =$ length of minimum length line, n in.
 D_n' = length of tangential segment of minimum length line, n in.
 $D_n'' =$ length of radial segment of minimum length line, n in.
 $E = \frac{\epsilon}{1-\epsilon}$ = emissivity parameter
 $F_{m-n} =$ view factor, fraction of isotropic radiation from A_m intercepted directly by A_n
 $\mathcal{F}_{m-n} =$ radiation exchange factor, fraction of radiation passing from A_m to A_n directly and indirectly
 $g =$ acceleration of gravity $\frac{\text{ft}}{\text{sec}^2}$
 $h_{\text{CON}} =$ convection heat transfer coefficient $\frac{\text{BTU}}{\text{hr-ft}^2 \cdot ^\circ F}$
 $h_{\text{RAD}} =$ radiation heat transfer coefficient $\frac{\text{BTU}}{\text{hr-ft}^2 \cdot ^\circ F}$
 $\bar{h} = h_{\text{CON}} + h_{\text{RAD}} =$ effective heat transfer coefficient $\frac{\text{BTU}}{\text{hr-ft}^2 \cdot ^\circ F}$
 $i = \sqrt{-1}$
 $J_n =$ radiosity of node n $\frac{\text{BTU}}{\text{hr-ft}^2}$
 $k =$ thermal conductivity $\frac{\text{BTU}}{\text{hr ft } ^\circ F}$

- k_c = effective thermal conductivity $\frac{\text{BTU}}{\text{hr ft}^\circ\text{F}}$
- r_n = radial distance from center of rocket motor to point n in
- r_o = inner radius of rocket motor in
- S_n = length of surface n in
- t = time min
- T = temperature of position r at time t $^\circ\text{R}$
- T_∞ = storage container temperature $^\circ\text{R}$
- T_M = maximum temperature of storage container $^\circ\text{R}$
- T_A = average temperature of storage container $^\circ\text{R}$
- $Z = \sqrt{\frac{i\omega r_o^2}{\alpha}} \xi$ = dimensionless distance parameter
- α = thermal diffusivity $\frac{\text{ft}^2}{\text{hr}}$
- $\beta = \frac{\bar{h}r_o}{k}$ = Biot modulus
- δ = width of air gap in
- ϵ = emissivity
- $\xi = \frac{r}{r_o}$ = dimensionless distance
- $\theta = \frac{T - T_A}{T_M - T_A}$ = dimensionless temperature
- θ^* = dimensionless temperature for supplementary problem
- θ_a = construction angle for crossed-strings method radians
- θ_r = relative amplitude of maximum temperature at point of interest to the maximum temperature of the storage container
- μ = dynamic viscosity $\frac{\text{lbm}}{\text{ft-hr}}$

- ρ = density $\frac{\text{lbm}}{\text{ft}^3}$
- σ = Stefan-Boltzman constant $0.171 \times 10^{-8} \frac{\text{BTU}}{\text{hr ft}^2 \cdot \text{R}^4}$
- τ = $\tau(t)$ = solution of ψ ;
 = $e^{im\omega t}$ for large values of time
- ϕ = $\phi(r)$ = solution of ψ
- ψ = complex temperature = $\theta^*(r,t) + i\theta(r,t)$
- ω = frequency of sinusoidal variation $\frac{2\pi}{24 \text{ hours}}$
- ω_T = resulting uncertainty in calculated temperature $^{\circ}\text{R}$
- ω_C = uncertainty in calculated temperature due to variation in volumetric heat capacity $^{\circ}\text{R}$
- ω_K = uncertainty in calculated temperature due to variation in conductivity $^{\circ}\text{R}$
- ω_ϵ = uncertainty in calculated temperature due to variation in emissivity $^{\circ}\text{R}$
- Gr = $\frac{\rho^2 g B (\Delta T) \delta^3}{\mu^2}$ = Grashof Number
- Pr = $\frac{c\mu}{k}$ = Prandtl Number

Bessel Functions

I_0 , J_0 , K_0 , BER , BEi

$$X_R = BER_0(a) + \frac{a}{\sqrt{2\beta}} BER_1(a) + \frac{a}{\sqrt{2\beta}} BEi_1(a)$$

$$X_i = BEi_0(a) + \frac{a}{\sqrt{2\beta}} BEi_1(a) - \frac{a}{\sqrt{2\beta}} BER_1(a)$$

$$\delta^* = \tan^{-1} \frac{BEi_0(a\xi)X_R - BER_0(a\xi)X_i}{BER_0(a\xi)X_R + BEi_0(a\xi)X_i} = \text{time delay} \quad \text{radians}$$

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I. INTRODUCTION

The purpose of this investigation was to develop a heat transfer model that will allow prediction of the temperature distribution in a container stored rocket motor placed in a hostile thermal environment such as the desert. It is proposed that such a model would be a useful tool for thermally optimizing future container designs. As extreme variations in the rocket motor temperature may lead to large thermal stresses in the propellant which could result in fracture, or otherwise degrade the performance of the motor, a major objective of this study was to design a model that could reliably predict the thermal gradient in the motor. The predictions would be based on the surface temperature distribution, the thermal properties and the geometrical details of the system. The model may also be used to predict a critical temperature range over which the propellant must be chemically stable when in a storage situation. The upper limit of this temperature range is referred to as the design temperature of the system. As the design temperature for most weapon development projects is derived from dump storage conditions, a dump storage situation was used to obtain the experimental data for this project.

Several approaches were taken to predict the rocket motor temperature distribution from a knowledge of only the surface temperature distribution of the storage container and the thermal properties and geometrical details of the

experimental model. The experimental model used in this test was a once-fired Navy antisubmarine rocket (ASROC) motor, filled with dry desert blow sand to simulate the propellant, and placed in its storage container. This container system was placed in a dump storage site at the Naval Weapons Center, China Lake, California to simulate a desert environment.

The method of complex temperatures [Ref. 1] was used to develop an analytical prediction of the transient temperature field that exists in a container stored rocket motor. The analytical model assumes that heat is transferred only in the radial direction and that the container surface temperature variation is sinusoidal with time. Comparison between theory and experiment is within experimental uncertainty when temperature is interpreted as "bulk" temperature. The analytical model is especially useful for studying geometrical and thermal physical property effects on rocket motor temperature. Such parameter studies have been carried out and the results are presented in a form that will be useful from a container design point of view.

TRUMP [Refs. 2 and 3], a computer program for transient and steady-state temperature distributions in multidimensional systems, was used to obtain detailed information about the thermal state of the rocket motor. TRUMP allows actual container surface temperature distributions to be used as well as sinusoidal variations. In addition, both one dimensional (radial) and two dimensional (radial and circumferential)

heat transfer were modeled with TRUMP, using both the sinusoidal and actual temperature distributions. The actual temperature distributions were obtained from the experimental data of the motor container system.

Comparisons between the experimental values and those predicted by the models were in good agreement, with those predicted by TRUMP using the actual temperature distribution as the boundary condition being the closest. However, the sinusoidal variations used in both the analytical model and the TRUMP model are also suitable for design purposes.

Another aspect of this project was to obtain the storage container surface temperature distribution using cholesteric liquid crystals, a material that undergoes brilliant changes in color over known, well defined temperature ranges. Color slides and movies were taken of the liquid crystals demonstrating the feasibility of using them for on site temperature measurements.

II. BACKGROUND

In 1959 the Naval Weapons Center, China Lake recognized the need for a concerted attack on the problem of thermal criteria assignment for new weapon systems. In 1963 a task force was established to study the complete environmental criteria determination problem. The key to this problem seemed to be the thermal area in the storage and transportation events of any item. It was realized that transportation was a short term situation compared to the storage situation. Therefore, the major portion of the life of an item must be in storage. There are three types of storage; covered, igloo and dump. The dump storage situation leads to the more extreme thermal exposure situations which then leads to the design temperature.

As data was not available for the dump storage situation, instrumented storage dumps were created at representative places on a worldwide basis so that statistical data could be derived on a variety of ordnance. The first site was at China Lake, California, in the middle of the Mojave Desert. This site now has the capability to return about 250 channels of information on a continuous time-temperature basis (Figure 1). Other arctic and tropical sites were set up to study extreme conditions.

The dump storage situation was reproduced to study the extreme situation. The ordnance was exposed singly, directly situated on the ground, with the long axis aligned in the



Figure 1. Simulated Storage Site at China Lake.

north-south direction to allow maximum normal exposure of the container surface to the sun's rays. In actual practice, ordnance is usually stacked and oriented in other than a north-south direction, thereby avoiding the extreme situation. Ordnance sitting on the ground receives reflected radiation from the ground, cannot quickly give off heat by conduction to the soil, and is not as apt to be cooled by the prevailing breeze; therefore, extreme temperatures result.

The most important source of heat to the ordnance is the direct radiation from the sun, with reflected radiation of secondary importance. For extreme conditions to occur the wind must be calm (less than 5 knots), the sky clear, and the outside air temperature high. After sunrise, the ordnance skin temperature rises much more rapidly than the ambient air temperature; therefore, the surrounding air cools the ordnance, rather than heats it.

The rocket motors used for the tests were military surplus. Even though the material had served its intended in-Fleet purpose, it was still representative of new hardware, when viewed in a thermodynamic context. When inert rocket motors were available, they were used intact; however, in most cases, once-fired hardware was used. Thoroughly dried desert blown sand, being similar in thermal properties to most propellants, was used to backfill empty rocket motors. It was assumed that the thermal response of the sand filled motors was essentially the same as actual propellant filled motors.

III. EXPERIMENTAL PROCEDURE

Although Naval Weapons Center, China Lake had accumulated vast amounts of data in the past, it was decided to instrument a rocket motor storage container system especially for this project. This would allow base data to be taken exactly where it was required. It also allowed variations in the system without interfering with one of China Lake's ongoing projects. An ASROC system was chosen for this study. The outer storage container was 75 inches long with an inner diameter of 18 inches and a wall thickness of 1/16 inch. The rocket motor was 57 inches long with an outside diameter of 12 inches and a wall thickness of 1/4 inch. Both the container and motor were made of steel.

The rocket motor storage container system was instrumented with 20 gage copper-constantan insulated thermocouple wire which has an ISA Calibration of $\pm 1-1/2^{\circ}\text{F}$ over the range -75 to $+200^{\circ}\text{F}$. Twenty-one thermocouples were originally placed on the system with positions indicated in Figures 2 and 3. The ambient air temperature was measured with thermocouple number 19 which was located in a Stevenson shelter about 60 feet away from the system (Figure 4). The thermocouples were mounted intrinsically on the motor and storage container. Two small holes were drilled approximately 1/8 inch apart in the metal and the individual wires were inserted in the holes. The metal was then hammered around the wires until a snug fit was obtained. Bead thermocouples were mounted at the

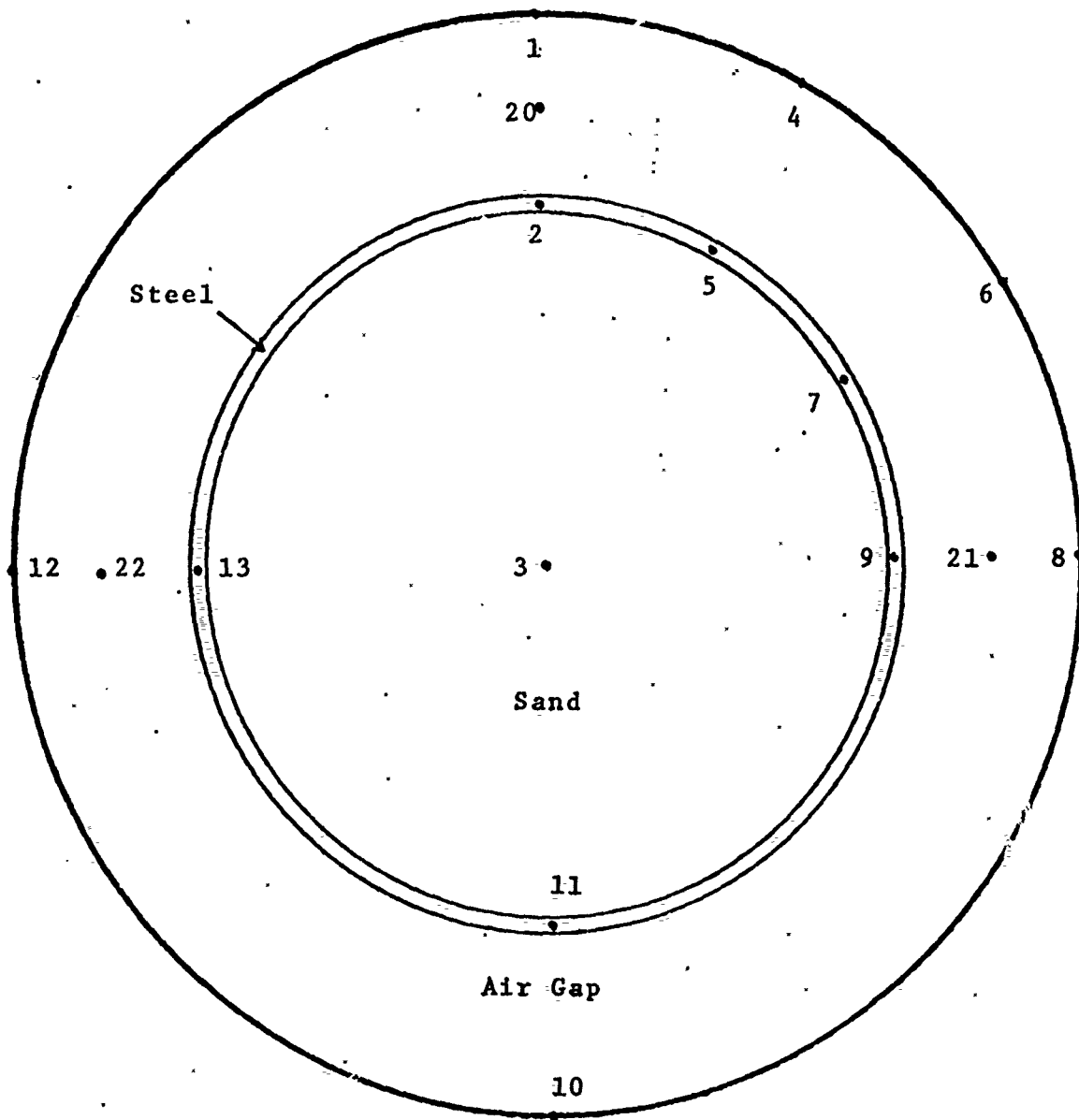


Figure 2. Thermocouple Locations on Experimental System.

Five thermocouples were located under the section painted with the liquid crystals. Their locations corresponding to the ones shown above are: #14= #1, #15= #2, #16= #8, #17= #9, and #18= #3 (See Figure 3).

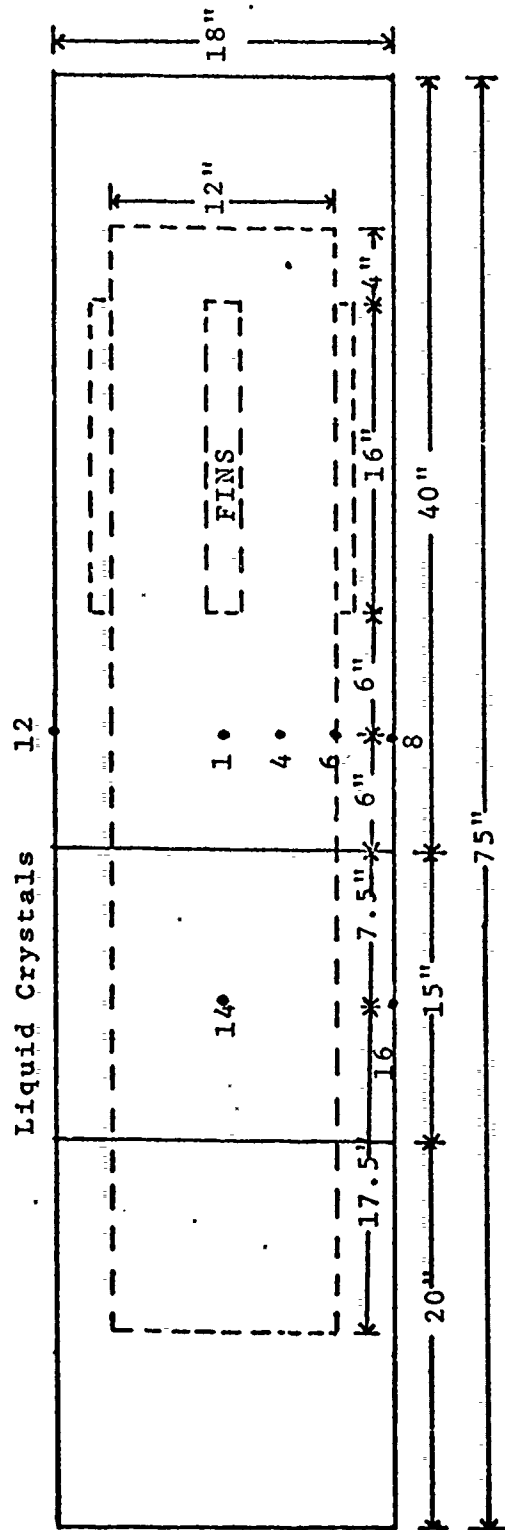


Figure 3. Top View of Rocket Motor Storage Container System.

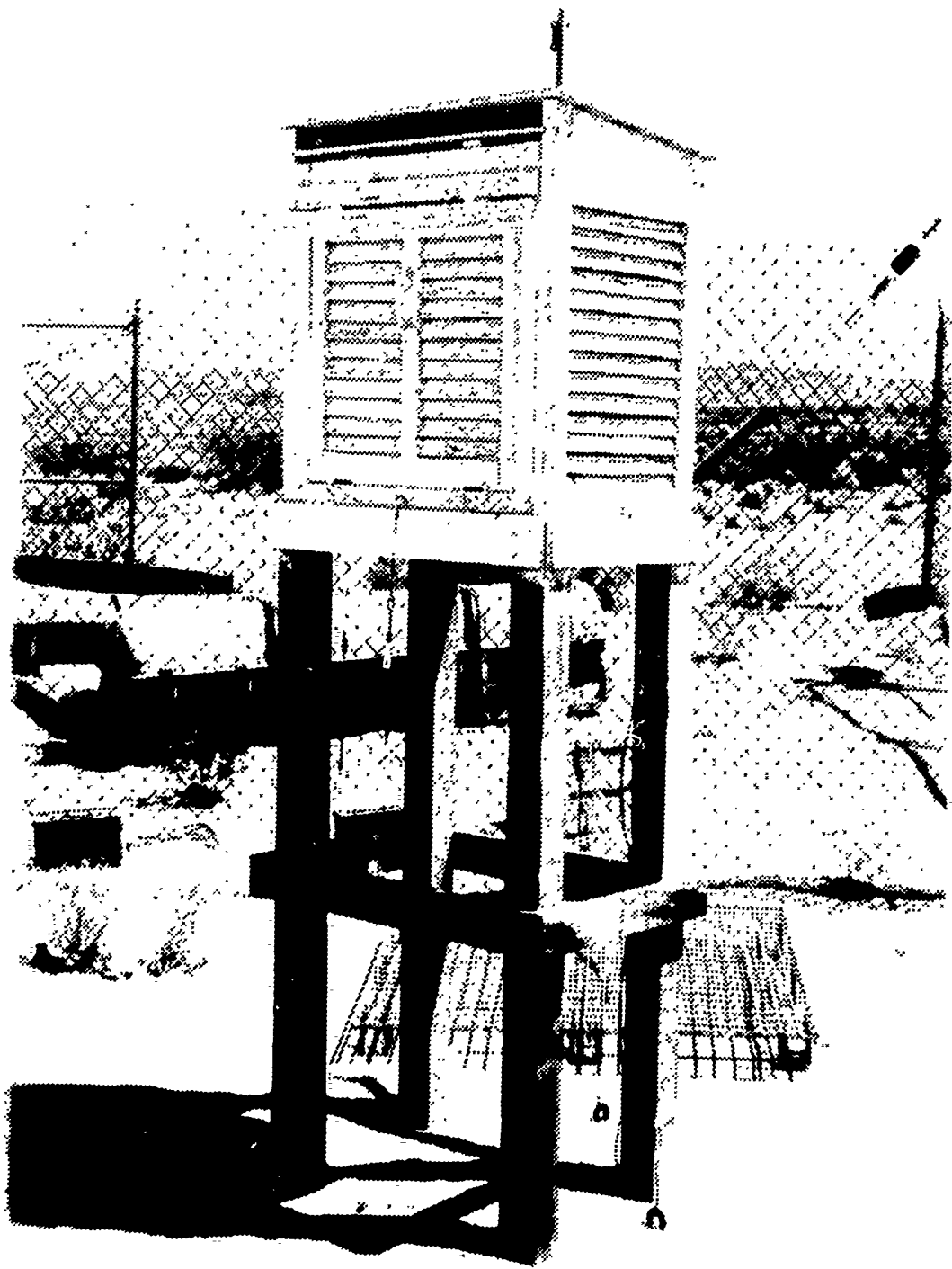


Figure 4. Stevenson Shelter.

center of the motor and in the air gap. The thermocouples located at the center of the motor were supported by small pieces of wood several inches from the head. The use of these supports was necessary to keep the thermocouples in position when the motor was being filled with sand. After all the thermocouples on the rocket motor were in place, the rocket motor was filled with dry desert blown sand. The wires from the two thermocouples located in the center of the motor were led out a hole in the end cap. To avoid settling of the sand after the motor was in place on the site, with a resulting air gap being formed between the sand and the motor skin, the sand was compacted by striking the sides of the motor with small sledge hammers and then adding additional sand through the hole in the end cap. This was continued until the sand was tightly packed. The hole in the end cap was then sealed. The rocket motor was carefully placed in its storage container (Figure 5) which had previously been instrumented with thermocouples. The thermocouples in the air gap were mounted by affixing the lead wire to the rocket motor at the desired position and then putting a 90 degree bend in the wire so that it placed the bead of the thermocouple approximately 1.5 inches into the air gap. Neither the thermocouples in the center of the motor nor those in the air gap could be considered accurately positioned; however, every effort was made to minimize positioning errors. All thermocouple wires were located inside the storage container and were led through a

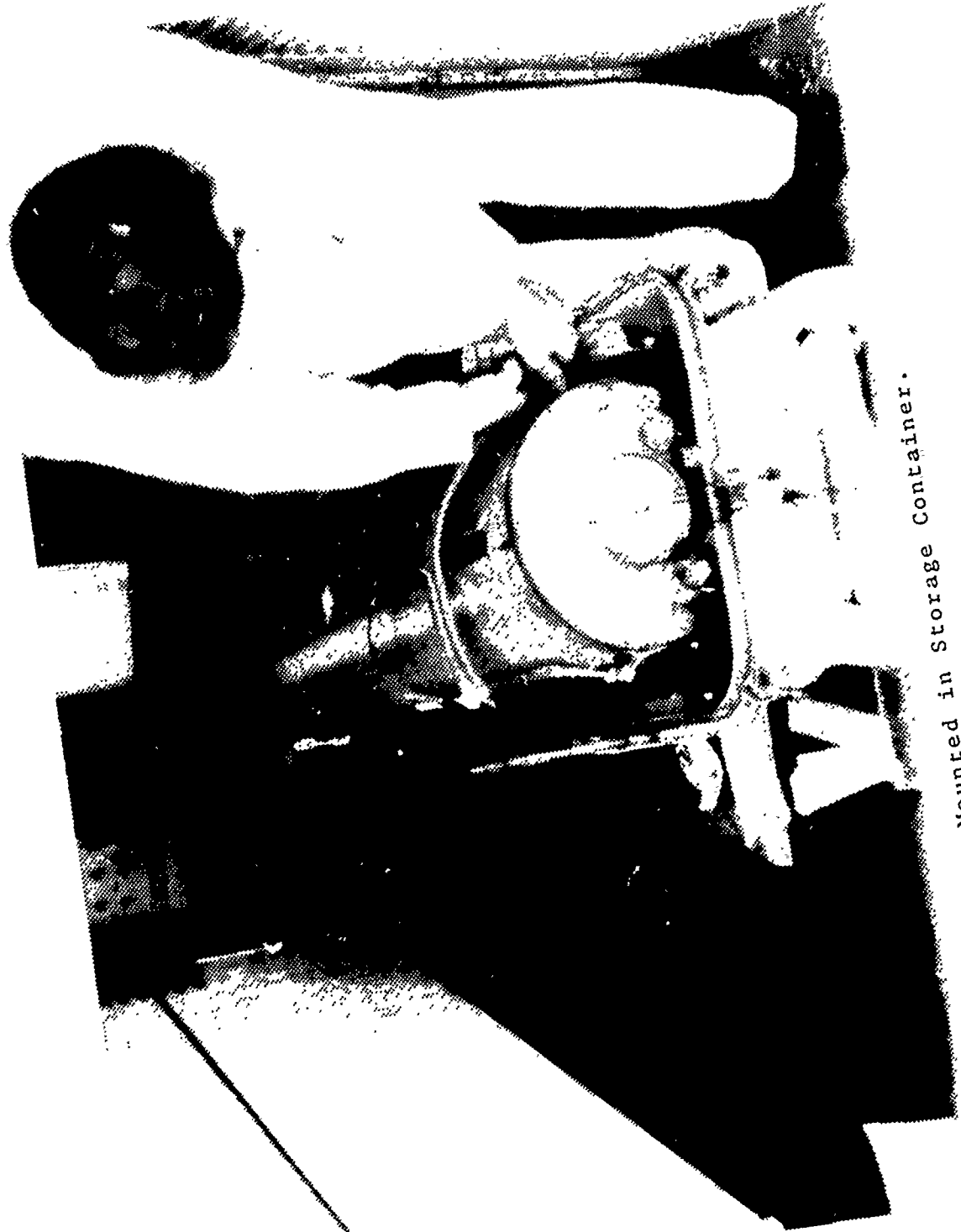



Figure 5. Rocket Motor Mounted in Storage Container.

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hole in one end. This hole was then sealed. The two halves of the storage container were then bolted shut.

The outer surface of the rocket motor and the inner and outer surfaces of the storage container were all painted various shades of haze gray. Weathering had caused the painted surfaces to appear fairly rough. This is typical of the conditions of a storage dump. From the condition of the surfaces, it was estimated that the emissivity was approximately 0.9.

Prior to loading the rocket motor into the storage container, it was decided to apply liquid crystals (See Appendix A) to part of the storage container surface in order to obtain a thermal mapping of the surface temperature at any instant of time. Liquid crystals are temperature sensitive materials that produce immediate thermal images in a pattern of colors which respond rapidly to minute changes in substrate surface temperatures. A second reason for applying the crystals to the container surface was to determine the feasibility of using the crystals under adverse environmental conditions (desert atmosphere). Prior to applying the crystals, a 15 inch strip of the storage container, 20 inches from one end, was sprayed with two coats of Testors Spray Pla Enamel No. 1249, Flat Black as a background for the crystals. A one inch strip of 11 different ranges of crystal, with approximately 1/2 inch of black paint between them, was applied over the black paint. Two coats of each crystal were applied, using a small paint brush. The first coat was allowed to dry completely before the second coat

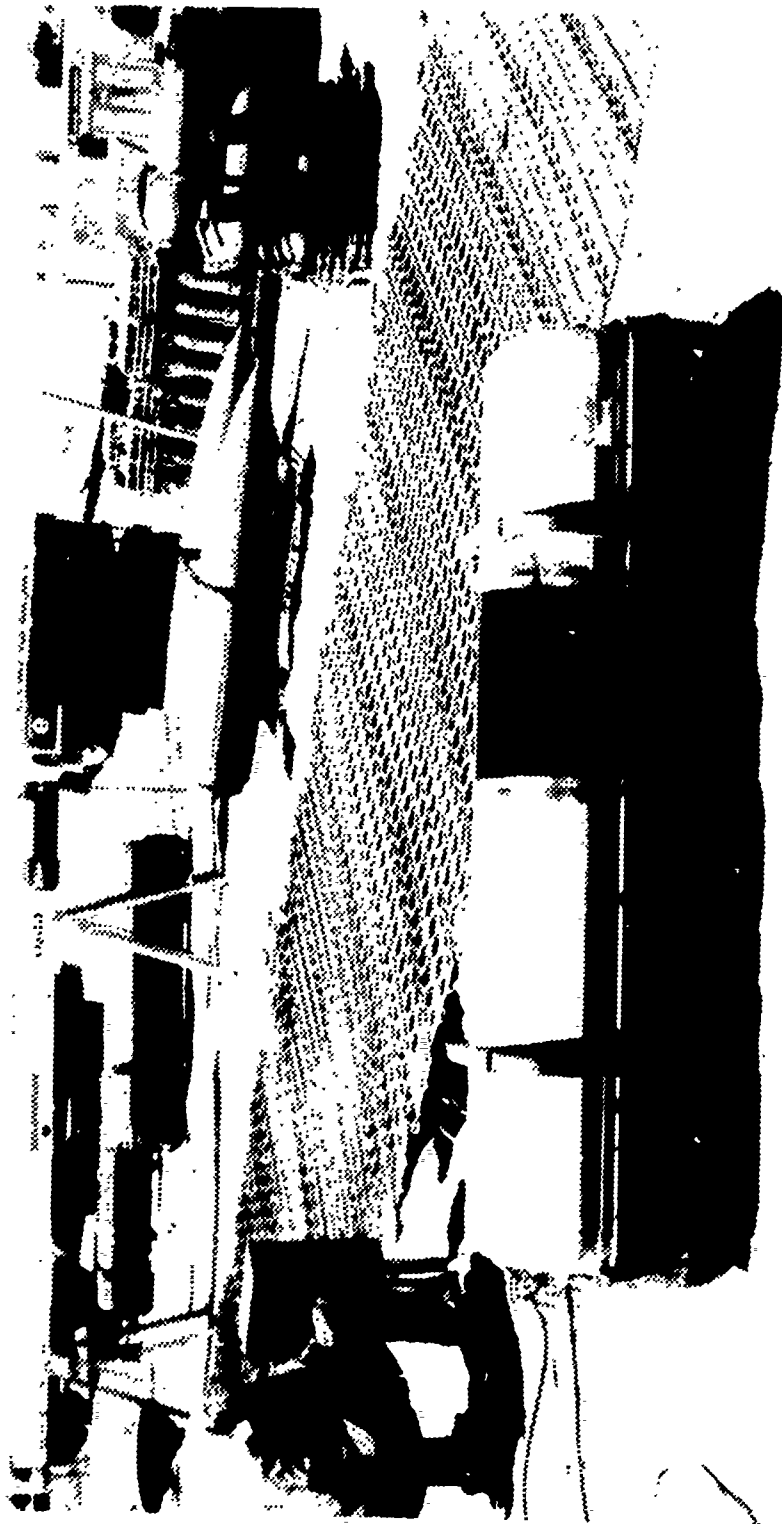
was applied. After the crystals were dry, two coats of Rez polyurethane (gloss clear plastic coating, interior-exterior 77-5) coating were applied by brush completely covering the crystals and black painted area. The polyurethane coating was applied to protect the crystals from wind blown sand and from the ultraviolet rays of the sun. Ten of the eleven crystals had been previously calibrated [Ref. 4]. Using the constant temperature bath procedure recommended in Ref. 4, R-27 was calibrated and the complete calibration results are shown in Table I.

The rocket motor storage container system was then moved to the China Lake dump storage site. The system was aligned in a north-south direction, well away from the influence of other ordnance (Figure 6). The thermocouple leads were connected through a junction box and underground cable to a Honeywell Electronik 25 Recorder which had been calibrated to read the thermocouple output directly in degrees Fahrenheit to an accuracy of $\pm 1^{\circ}\text{F}$. The recorder was located in an air-conditioned building about 60 feet from the system.

Initial data indicated that the number 7 thermocouple was not responding properly and therefore this data was neglected. Initial color photographs were taken of the liquid crystals and it was immediately apparent that good thermal mappings could be obtained if the crystals were stable under the adverse desert environment. The brilliance of the colors exhibited by the crystals under the bright desert sun was much better than had been expected. The

TABLE I
Calibration of Liquid Crystals

| NCR Desig. | Color Change | Manufacturer's | Calibration Bath |
|---------------|--------------|----------------|----------------------------|
| | | Responses | 2 Coats Liquid Crystals |
| | | °C | °C |
| R-27 | Red | 27.0 | 25.6 \pm .5 |
| | Green | 28.6 | 28.0 \pm .5 |
| | Blue | 30.0 | 28.7 \pm .5 |
| R-33 | Red | 33.0 | 32.7 \pm .5 |
| | Green | 34.6 | 33.3 \pm .5 |
| | Blue | 36.0 | 34.2 \pm .5 |
| R-37 | Red | 37.0 | 36.2 \pm .5 |
| | Green | 38.6 | 37.1 \pm .5 |
| | Blue | 40.0 | 38.0 \pm .5 |
| R-41 | Red | 41.0 | 40.3 \pm .5 |
| | Green | 42.6 | 41.0 \pm .5 |
| | Blue | 44.0 | 42.0 \pm .5 |
| R-45 | Red | 45.0 | 42.8 \pm .5 |
| | Green | 46.6 | 43.6 \pm .5 |
| | Blue | 48.0 | 44.3 \pm .5 |
| R-49 | Red | 49.0 | 46.7 \pm .5 |
| | Green | 50.6 | 47.1 \pm .5 |
| | Blue | 52.0 | 48.4 \pm .5 |
| R-53 | Red | 53.0 | 50.5 \pm .5 |
| | Green | 54.6 | 52.1 \pm .5 |
| | Blue | 56.0 | 53.3 \pm .5 |
| R-56 | Red | 56.0 | 53.8 \pm .5 |
| | Green | 57.6 | 56.0 \pm .5 |
| | Blue | 59.0 | 56.5 \pm .5 |
| R-59 | Red | 59.0 | 56.9 \pm .5 |
| | Green | 60.6 | 57.5 \pm .5 |
| | Blue | 62.0 | 58.9 \pm .5 |
| S-62 | Red | 62.0 | 60.1 \pm .5 |
| | Green | 62.6 | 60.4 \pm .5 |
| | Blue | 63.0 | 60.9 \pm .5 |
| S-64 | Red | 64.0 | 60.9 \pm .5 |
| | Green | 64.6 | 61.4 \pm .5 |
| | Blue | 65.0 | 62.7 \pm .5 |




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Figure 6. Experimental System at Dump Storage Site.

system was allowed two weeks to reach a periodic steady state before additional photographic data was obtained. . .

Extensive photographic data was collected on 27 and 28 July 1972 after two weeks of exposure to the desert environment. Both super 8 mm and 16 mm color movies and 35 mm color slides were taken of the liquid crystals. No colored filters were used on any of the cameras, although standard haze filters were used to take the super 8 mm movies and most of the 35 mm slides.

At this time, a second storage container, this one without a rocket motor inside, was instrumented with intrinsic thermocouples in the same manner as the previous container. As only three data channels remained open on the recorder, only three thermocouples were applied to this new container. The three thermocouples were applied at the 0300, 0900, and 1200 positions at the midpoint of the container. This container was set end to end with the system that was already in place at the site. The purpose of this study was to determine if the inclusion of the rocket motor in the container had a significant effect on the surface temperature of the container. Thermocouple #7 was connected at the 0900 position, #23 at the 1200 position, and #24 at the 0300 position. It was immediately apparent that thermocouple #7 was continuing to give unreliable readings and therefore the data taken on channel #7 was neglected.

IV. THEORETICAL ANALYSIS

A. ONE-DIMENSIONAL ANALYTICAL MODEL

The first step was to try to devise an analytical model that would simulate the actual rocket motor storage container experimental system. The first simplifying assumption was that the storage container temperature could be modeled by a sine wave which had a period of 24 hours. A comparison of the sinusoidal variation to the average (bulk) storage container temperature [obtained by averaging the four thermocouple readings on the surface of the container (1, 8, 10, and 12) as shown in Appendix D] is given in Figure 7.

The method of complex temperature as presented by Arpaci [Ref. 1] was used to find the steady periodic solution of a body experiencing a periodic sinusoidal disturbance. A complete analytical derivation is given in Appendix B. The general heat conduction equation in cylindrical coordinates was the basis for this derivation. It was assumed that there was one dimensional radial heat flow with no conduction in the axial or circumferential directions, that no heat sources existed in the model, that the rocket motor storage container system was infinitely long, and that the sinusoidally varying surface temperature was spatially uniform over the entire container surface. The storage container temperature is assumed to vary as

$$T_{\infty} = (T_M - T_A) \sin \omega t + T_A$$

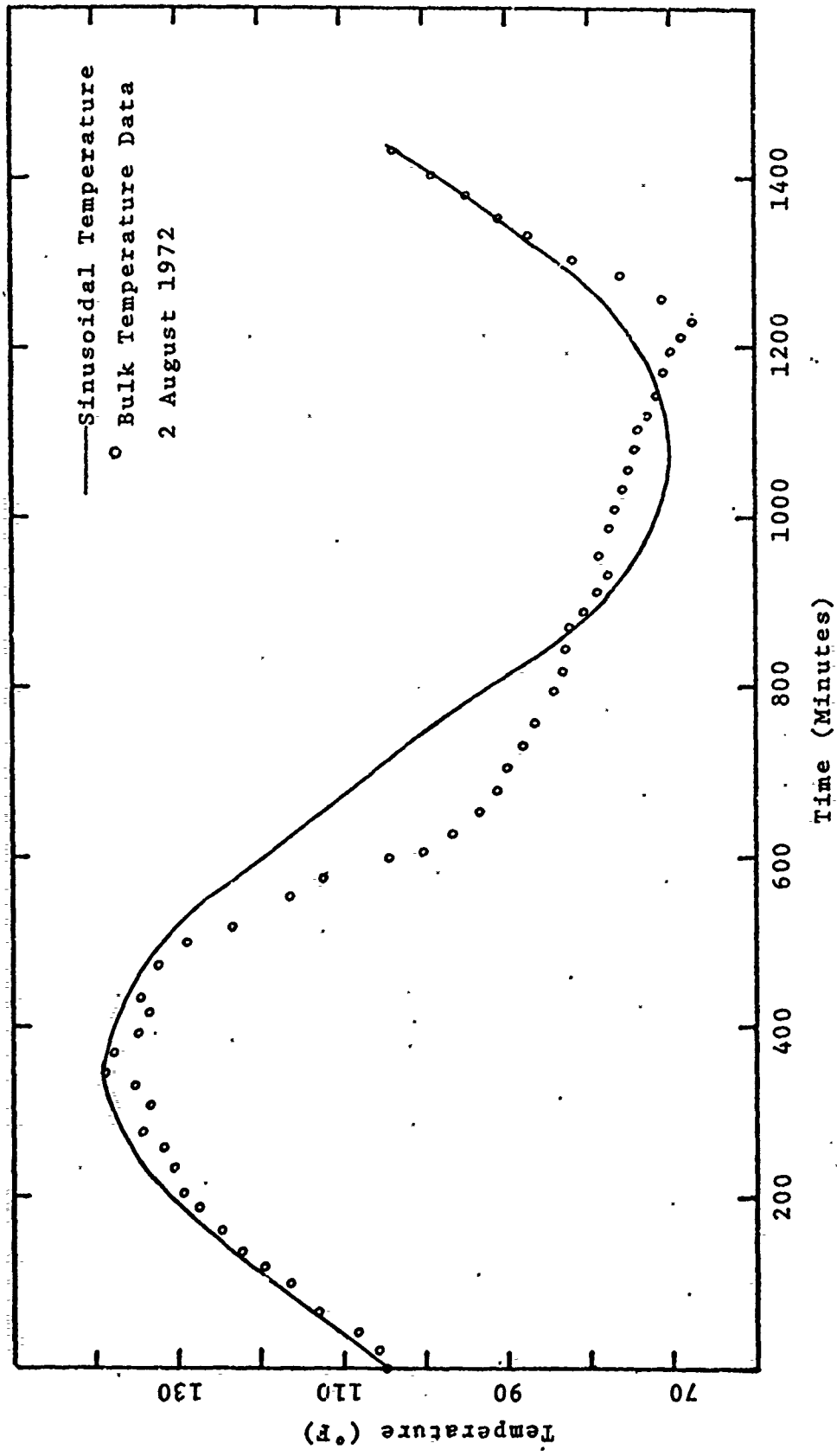


Figure 7. Comparison of Sinusoidal, Temperature Variation to Bulk Temperature.

where T_M = maximum bulk temperature of the storage container
 T_A = average bulk temperature of the storage container
 ω = frequency of the sinusoidal variation
 (2 π /24 hours)
 t = time (hours)

It was assumed that all the thermal properties remained constant over the temperature range of the problem. The effective heat transfer coefficient, \bar{h} , across the air gap between the storage container and the rocket motor combines the heat transfer effects of radiation, convection, and conduction into one coefficient. The radiation coefficient was linearized by assuming constant representative temperatures in the equation

$$h_{RAD} = \mathcal{F}_{1-2} \sigma (T_1 + T_2)(T_1^2 + T_2^2)$$

where σ is the Stefan-Boltzmann constant and \mathcal{F}_{1-2} is the radiation exchange factor. The convection coefficient is the effective conductivity of air, obtained from the Beckmann correlations [Ref. 5], divided by the width of the air gap. In the analytical model, the effective conductivity was assumed to equal the conductivity, thereby treating it as pure conduction and giving the equation

$$\bar{h} = h_{RAD} + h_{CON}$$

An initial condition was not specified in this derivation as the only concern was the steady-state, periodic behavior. The steady-state solution is (Appendix B)

$$\theta(r,t) = \frac{T(r,t) - T_A}{T_M - T_A} = \frac{\sqrt{BER_o^2(a\xi) + BEi_o^2(a\xi)}}{\sqrt{X_R^2 + X_I^2}} \sin(\omega t + \delta^*)$$

$$\theta(r,t) = \theta_r \sin(\omega t + \delta^*)$$

where $T(r,t)$ = the temperature of a point r in the rocket motor at time t

$$a = \sqrt{\frac{\omega r_o^2}{\alpha}} = \text{conduction parameter}$$

$$\xi = \frac{r}{r_o} = \text{dimensionless distance from the center of the rocket motor}$$

r_o = inner radius of the rocket motor

r = distance from the center of the rocket motor

$$\alpha = \frac{k}{\rho c} = \text{thermal diffusivity}$$

ρ = density

k = thermal conductivity

c = specific heat

BER = real Bessel Function

BEi = imaginary Bessel Function

$$X_R = \text{BER}_o(a) + \frac{a}{\sqrt{2}\beta} \text{BER}_1(a) + \frac{a}{\sqrt{2}\beta} \text{BEi}_1(a)$$

$$X_i = \text{BEi}_o(a) + \frac{a}{\sqrt{2}\beta} \text{BEi}_1(a) - \frac{a}{\sqrt{2}\beta} \text{BER}_1(a)$$

$$\beta = \frac{\bar{h}r_o}{k} = \text{Biot modulus}$$

$$\delta^* = \tan^{-1} \frac{\text{BEi}_o(a\xi)X_R - \text{BER}_o(a\xi)X_i}{\text{BER}_o(a\xi)X_R + \text{BEi}_o(a\xi)X_i}$$

Two computer studies were done based on the steady state solution. The first study was a completely dimensionless situation which served as a parameter study of the effects of varying a and β on the temperature and the time lag of the temperature at various positions in the model.

$$a = \sqrt{\frac{\omega r_o^2}{\alpha}} = \text{conduction parameter}$$

and

$$\beta = \frac{\bar{h}r_o}{k} = \text{Biot modulus}$$

Parameter a was varied from 1.0 to 5.0 and β was varied from 0.1 to 100. These were the only values studied, as

only values within this range are of interest in this type problem. The computer program and its output are given at the end of Appendix B. The output lists the following values:

- 1) a , the conduction parameter
- 2) β , the Biot modulus
- 3) ξ , the non-dimensional distance from the center of the motor
- 4) δ^* , the time delay between the maximum storage container temperature and the maximum temperature reached at the point of interest in the motor
- 5) θ_r , the relative amplitude of the maximum temperature at the point of interest compared to the maximum temperature of the storage container

The time delay is given in radians, where 2π radians equals one complete cycle. A graph of the time delay versus β for a constant value of " a " is given in Figure 8 at three different positions within the motor. A graph of the relative amplitudes of the temperatures versus β for a constant value of " a " is given in Figure 9. It was noted that for a constant value of " a ", the time delay decreased as β became larger. As the point of interest approaches the center of the rocket motor, the time delay increases. The relative amplitude of the temperatures also becomes larger as β is increased when the value of " a " is held constant. If β is held constant and " a " is varied, the time delay increases and the relative amplitude decreases as " a " increases.

The second study was obtaining the analytical solution to the particular rocket motor storage container system.

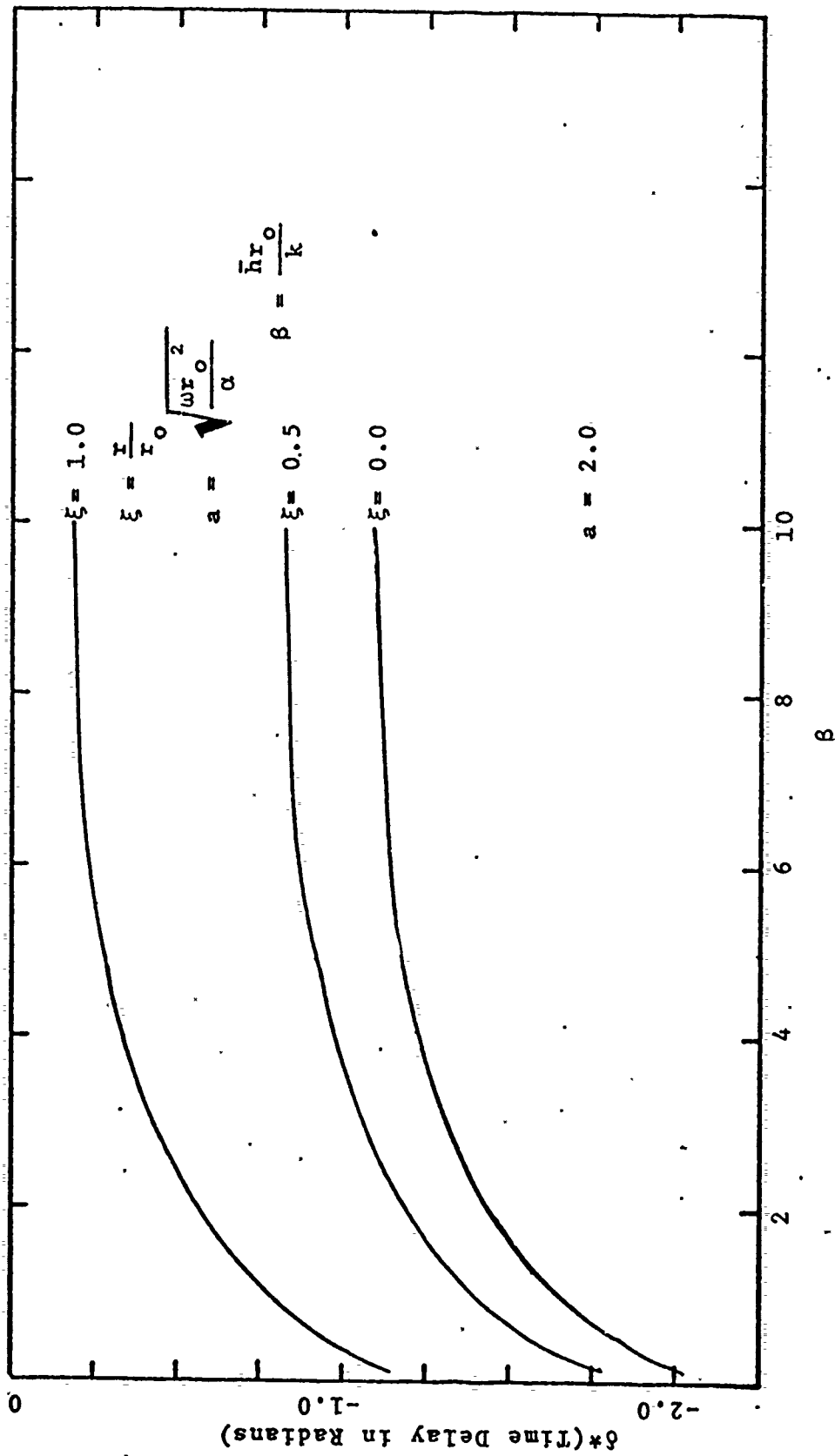


Figure 8. Variation in Time Delay with Change in Biot Modulus

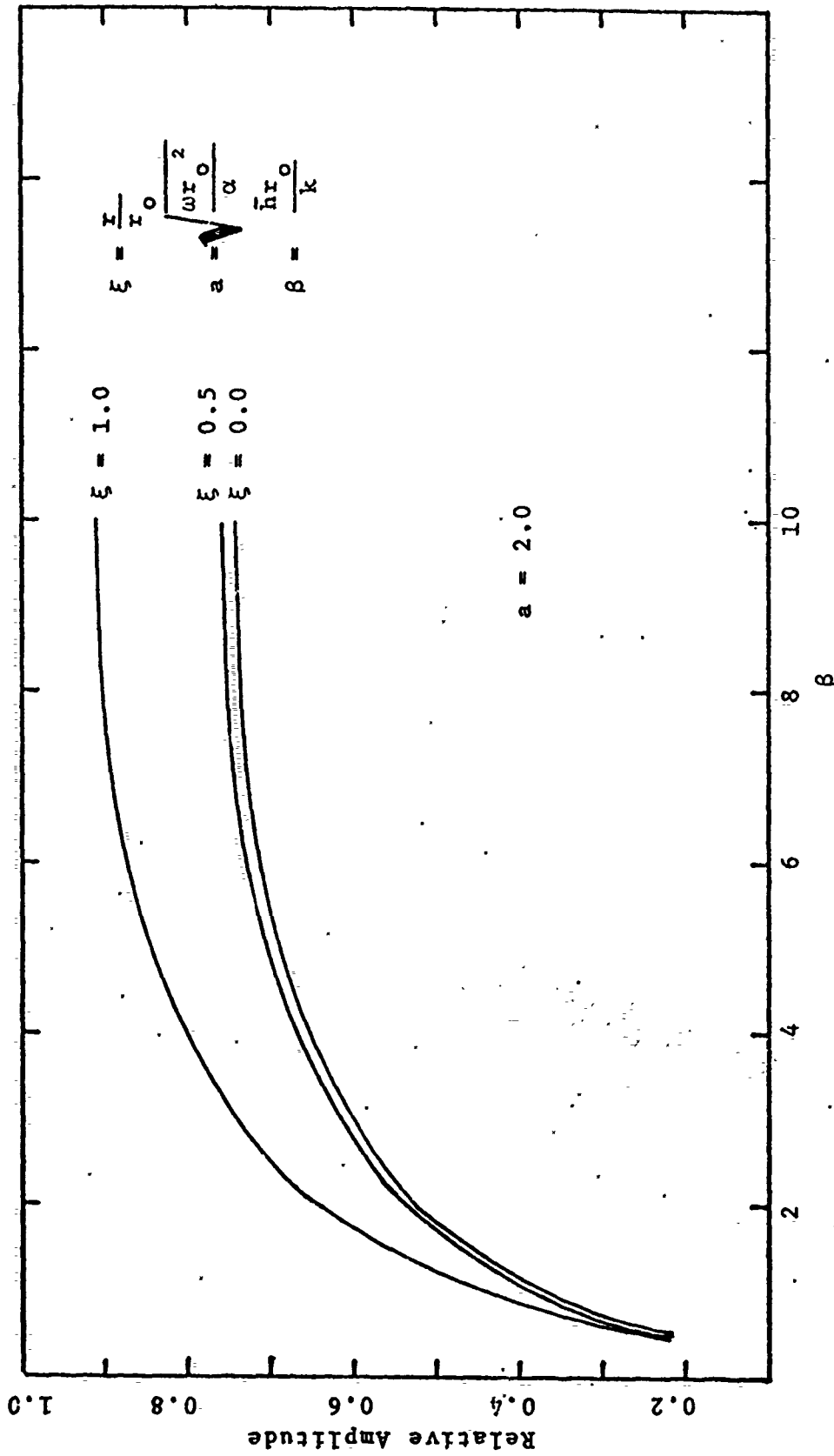


Figure 9: Variation in Relative Amplitude with Change in Biot Modulus.

studied at China Lake. The thermodynamic properties of dry sand were obtained from Ref. 6 as

$$\begin{aligned}\rho &= 94.8 \text{ lbm/ft}^3 \\ k &= 0.188 \text{ BTU/hr ft } ^\circ\text{F} \\ c &= 0.195 \text{ BTU/lbm } ^\circ\text{F}\end{aligned}$$

Substituting these values and using 1440 minutes (24 hours) as a complete cycle, the parameters a and β were calculated for this model as

$$a = \sqrt{\frac{\omega r_o^2}{\alpha}} = 2.43$$

where $r_o = 5.75$ inches, the inner radius of the rocket motor.

$$\beta = \frac{\bar{h} r_o}{k} = 2.90$$

where $\bar{h} = h_{\text{CON}} + h_{\text{RAD}}$

$$\text{and } h_{\text{CON}} = \frac{k_{\text{AIR}}}{\Delta r} = 6.48 \times 10^{-2} \frac{\text{BTU}}{\text{hr-ft}^2\text{ } ^\circ\text{F}}$$

where $\Delta r = 2.94$ inches, the distance across the air gap

$$\text{and } k_{\text{AIR}} = 1.62 \times 10^{-2} \frac{\text{BTU}}{\text{hr ft } ^\circ\text{F}}$$

$$h_{\text{RAD}} = \mathcal{F}_{1-2} \sigma (T_1 + T_2) (T_1^2 + T_2^2) = 1.09 \frac{\text{BTU}}{\text{hr ft}^2\text{ } ^\circ\text{F}}$$

where σ is the Stefan-Boltzmann constant, \mathcal{F}_{1-2} is the radiation exchange factor which for this geometry is

$$\mathcal{F}_{1-2} = \frac{1}{\frac{1}{\epsilon_1} + \frac{r_1}{r_2} \left(\frac{1}{\epsilon_2} - 1 \right)} = 0.84$$

when $\epsilon_1 = \epsilon_2 = .9$, $r_1 = 6.0$, $r_2 = 8.94$

therefore $\bar{h} = 1.15 \text{ BTU/hr ft}^2\text{ } ^\circ\text{F}$

The average surface temperature of the storage container was found to be 104°F for a particular day at China Lake,

with a maximum temperature of 138^oF. These values were obtained by averaging the readings of thermocouples 1, 8, 10, and 12 as shown in Appendix D which give the bulk temperature.

The temperatures of seven positions within the rocket motor were calculated and the results are printed at 30 minute intervals for one complete cycle in Appendix B. A graph of temperature versus time was plotted by the computer showing the relationship between the surface temperature of the storage container (TINF), the temperature on the outer skin of the rocket motor (TEDG), and the temperature at the center of the motor (TCEN). This graph is Figure 10.

B. TRUMP MODEL

The rocket motor storage container system at China Lake was modeled on TRUMP, a numerical conduction code, (See Appendix C for a description of the TRUMP program) to predict the temperature at any point in the system from a knowledge of the storage container surface temperature variation, the thermal properties and the geometrical details of the system. Two models were used to simulate the rocket motor storage container system and several variations of each model were investigated.

The first model assumed one dimensional heat transfer (radial). The system was modeled as two infinitely long concentric steel cylinders, the inner of which was filled with dry sand. A 2.94 inch air gap separated the cylinders. The model was subdivided into concentric volumetric elements

with representative nodal points as given in Figure 27, Appendix C. It was assumed that the storage container surface temperature was spacially uniform. From the data given in Appendix D and the observation of the liquid crystals' thermal mapping, it was obvious that the temperature distribution on the storage container was not spacially uniform. In order to simulate a spacially uniform condition, the readings of the thermocouples located at the 1200, 0300, 0600, and 0900 positions (#1, 8, 10, and 12) were averaged and this average value of the surface temperature (referred to as the bulk temperature) was used as the spacially uniform temperature distribution. Two methods were used to describe the container temperature. The first method used the maximum bulk temperature (138°F) and the average bulk temperature (104°F) of the storage container to generate a sine wave with a period of 24 hours (1440 minutes). The second method took the bulk temperature readings at two hour intervals and fed this data into the TRUMP program in a tabular (temperature versus time) form. The version of TRUMP used in this problem was limited to a table length of 12 tabular values. TRUMP interpolated between the tabular points. Figure 11 compares the actual bulk data with the sinusoidal approximation and the interpolated tabular values.

Several assumptions were made to simplify the solution of this problem. As the thermocouple data from the storage container gave an average value of the temperature across the 1/16 inch steel wall, node 12 was modeled as a zero

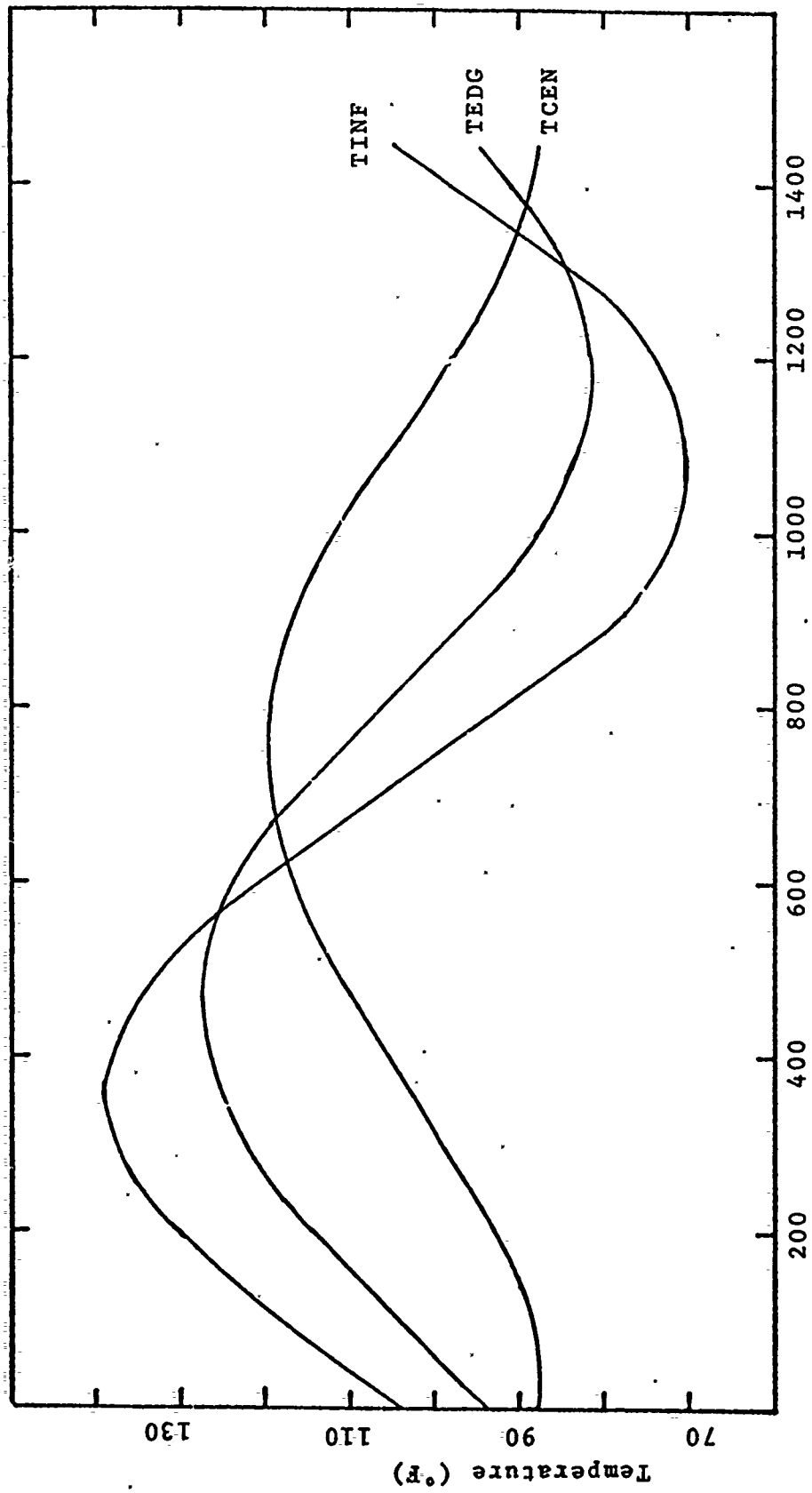


Figure 10: Analytical Prediction of Temperature Variation with Time, where TINF is the surface temperature of the storage container, TEDG is the surface temperature of the rocket motor, and TCEN is the temperature at the center of the motor.

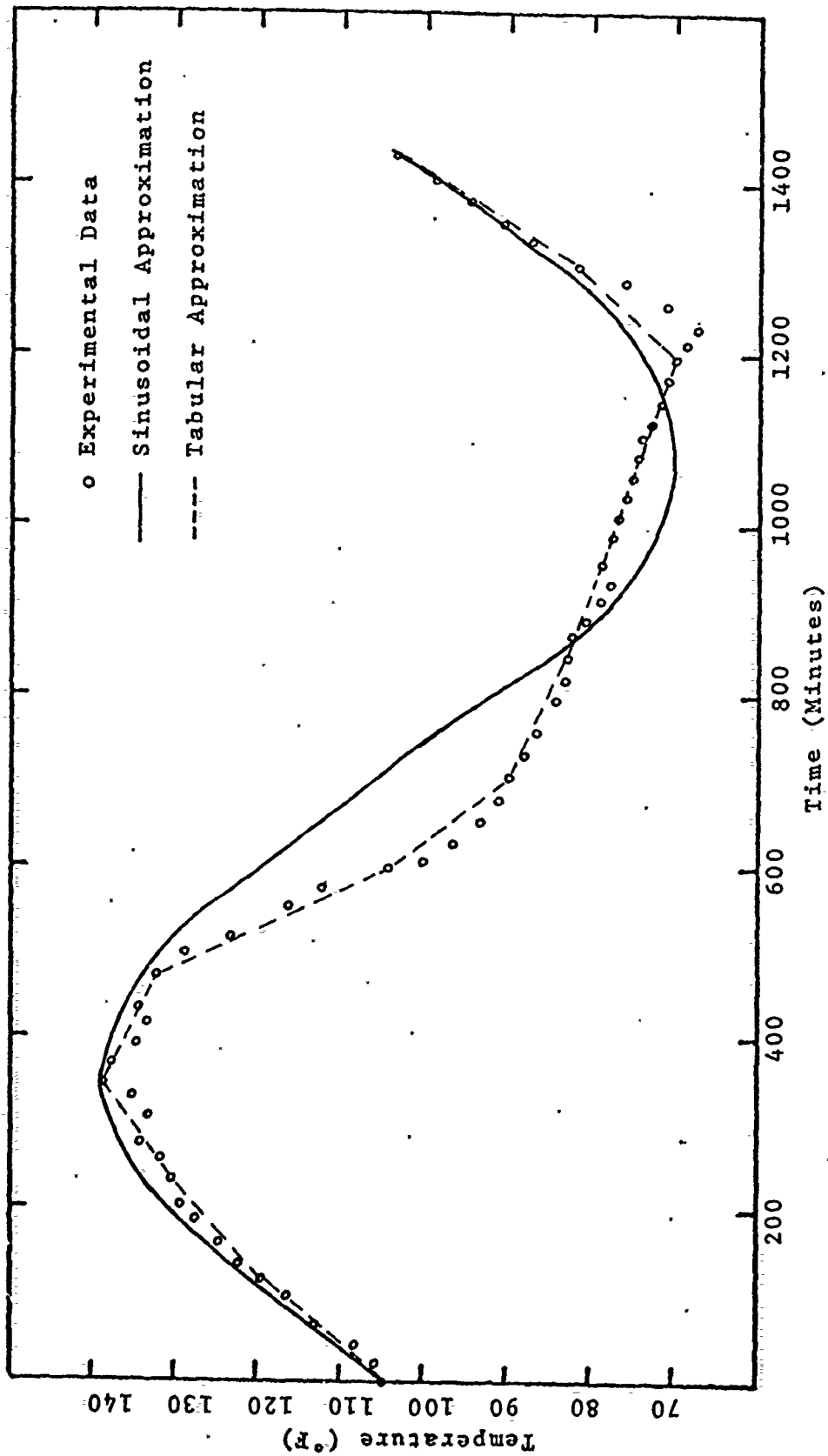


Figure 11: Comparison of Bulk Temperature to Two TRUMP Approximations.

volume boundary node with a known temperature impressed upon it. It was also assumed that heat transfer across the air gap occurred by radiation and conduction alone. Free convection effects were initially neglected. This assumption was later modified to investigate the free convection effects. All surfaces of the storage container and the outside surface of the rocket motor were painted various shades of haze gray and it was estimated that the emissivity of these surfaces was 0.9. The radiation exchange factor, F_{1-2} , for this model was the same as that for the analytical solution ($F_{1-2} = 0.84$). It was also assumed that there was perfect thermal contact between the rocket motor and the sand that filled it. This neglects the possibility that the sand might slightly settle after being on the site for a long period of time.

The second model assumed two dimensional heat transfer (radial and circumferential). The same physical model was used as in the one dimensional case with the sole exception that 48 nodes were used instead of 12. The representative nodal points and an example of the thermal connections from one of the nodal points are shown in Figure 28 in Appendix C. The four nodes on the surface of the storage container were modeled as zero volume boundary nodes. The sinusoidal and tabular representations were used to describe the surface temperature of the storage container at each boundary node. Actual data taken at each position, rather than bulk data, were used as the input data for these representations.

The same assumptions made in the one dimensional case were also applicable to the two dimensional model. A complete discussion of the calculation of the radiation exchange factors in the two dimensional case is given in Appendix C.

The effect of natural convection was studied in both the one and two dimensional models. References 5 and 7 give correlations between the Grashof number based on the gap width and the effective thermal conductivity. The Grashof number was calculated from the equation

$$Gr = \frac{\rho^2 g B (\Delta T) \delta^3}{\mu^2}$$

where δ = width of the air gap

ΔT = maximum temperature difference at any instant of time in the air gap

$B = \frac{1}{T}$ where $T = 565^\circ R$

At $T = 565^\circ R$, air has the following properties

$$\rho = 0.07 \text{ lbm/ft}^3$$

$$\mu = 0.046 \text{ lbm/hr-ft}$$

The maximum Grashof number for this experiment was calculated to be 1.25×10^6 . The diameter ratio was approximately 1.5 and the $\log Gr = 6.1$. From the Beckmann correlation [Ref. 5], the effective thermal conductivity ratio $\left(\frac{k_c}{k}\right)$ was approximately 3.2. Using the Liu correlation [Ref. 7]

$$\frac{k_c}{k} = 0.135 \left(\frac{Pr^2 Gr \delta}{1.36 + Pr} \right)^{0.278} = 4.5$$

where the Prandtl Number = 0.707. An effective thermal conductivity of 4.0 was assumed as the average value of these

two correlations and it was used to study the effects of free convection. This change was placed into the TRUMP program by increasing the value of the thermal conductivity of air by a factor of 4 in each of the TRUMP runs.

V. RESULTS

A. ANALYTICAL MODEL

Using the sinusoidal temperature distribution as an approximation to the actual average experimental data as shown in Figure 7, comparisons were made between predicted temperatures and actual temperatures for two radial locations in the rocket motor. Figure 12 compares the results on the surface of the rocket motor and Figure 13 does the comparison at the center of the rocket motor. An uncertainty analysis is given in Appendix E which establishes the uncertainty bounds for both the predicted and the actual temperatures. These uncertainty bounds are included in Figures 12 and 13.

It is readily seen from Figure 7 that a sine wave was not an ideal fit as an approximation to the experimental data, as it varies as much as 20°F during part of the cycle. However, it was also noted that the sine wave closely approximated the experimental data during the heating phase of the cycle and only during the cooling phase were there large variations. As the main purpose of this study was to design a model that would be useful in optimizing storage container design, the errors in the cooling phase are not critical as long as the temperatures reach the same minimum point before beginning another cycle. Figure 12 shows that the maximum surface temperature of the rocket motor predicted by the analytical model is a good approximation to the actual

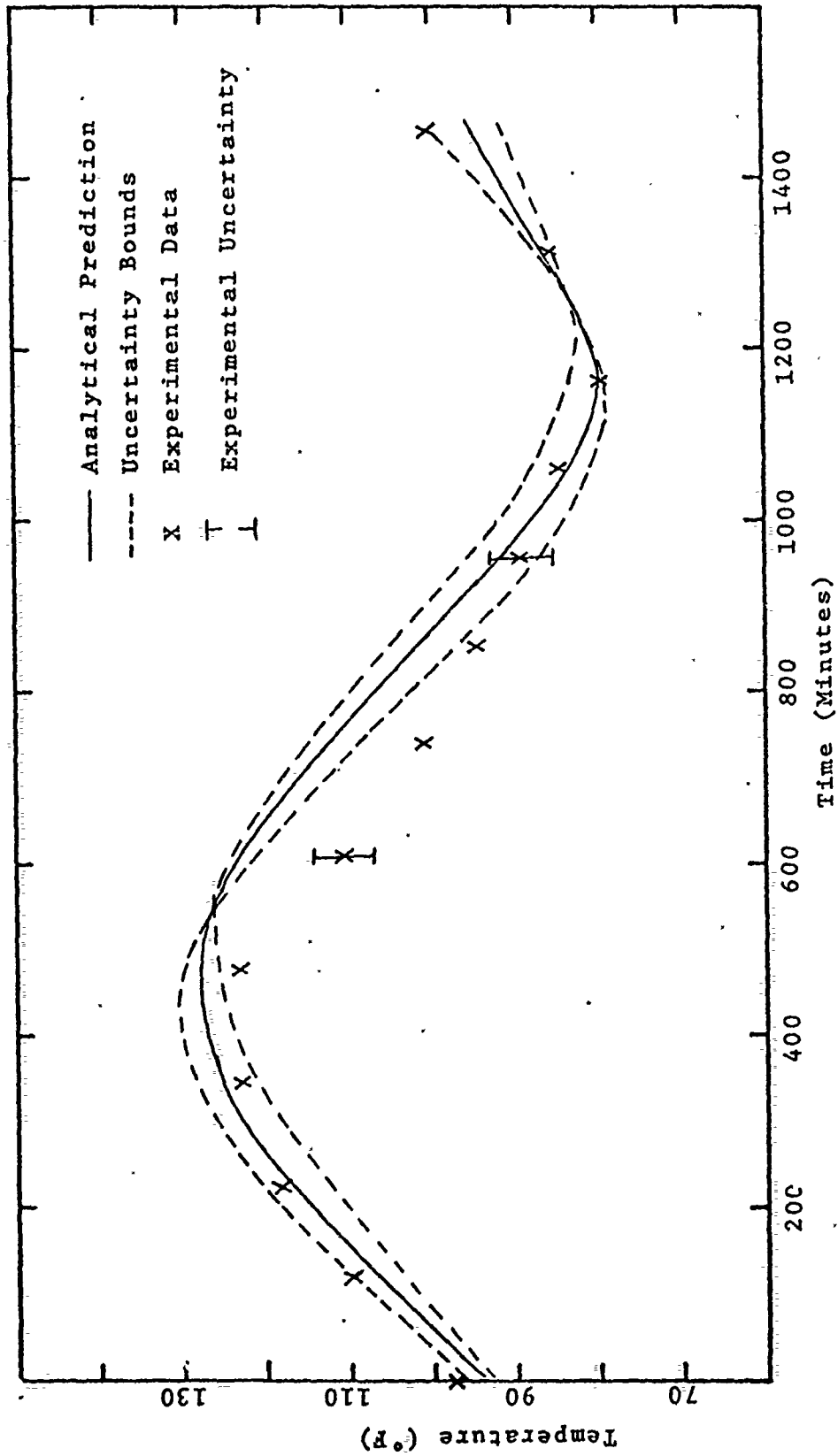


Figure 12: Comparison of Analytical and Experimental Temperatures at Surface of Rocket Motor.

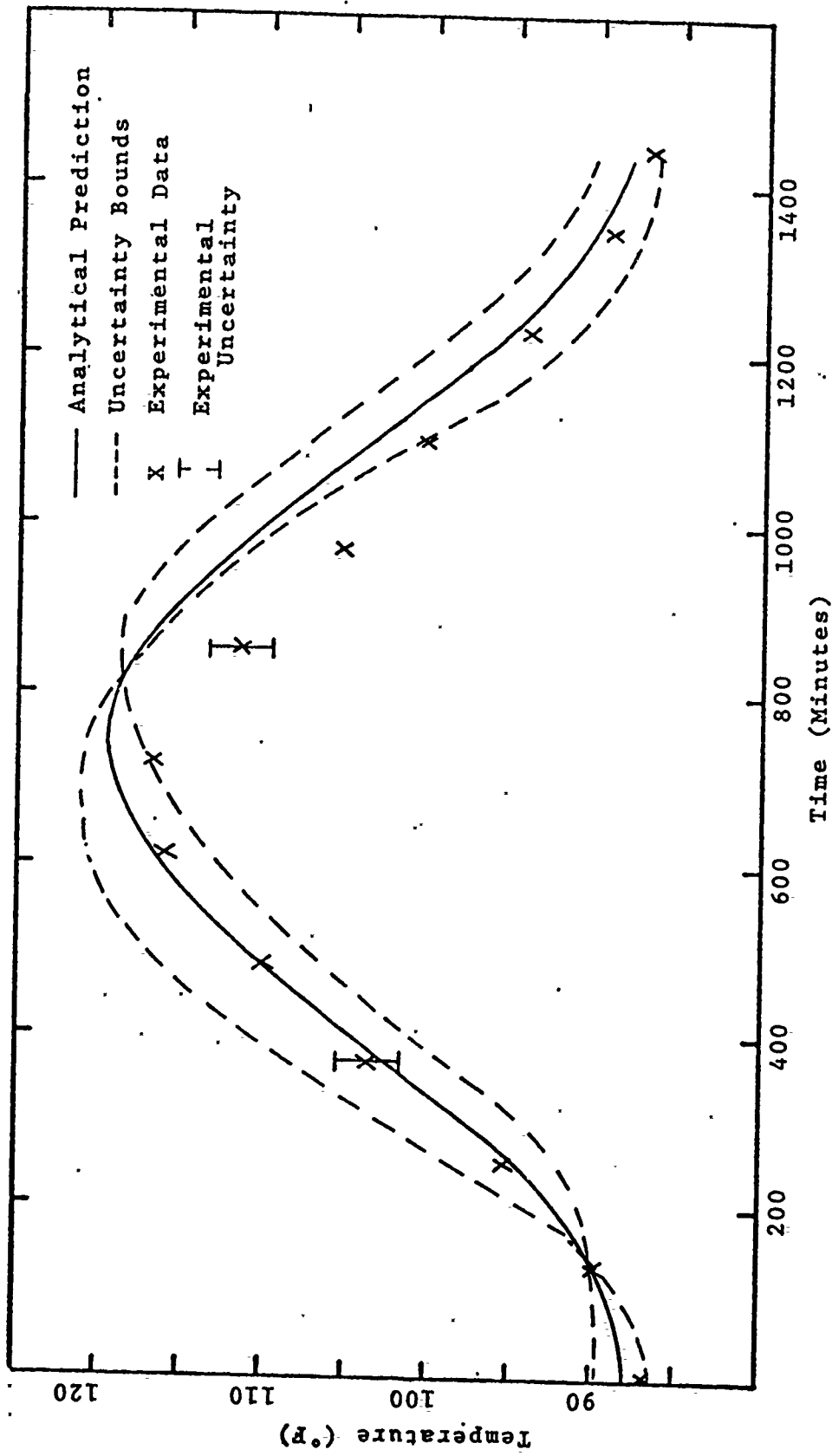


Figure 13: Comparison of Analytical and Experimental Temperature at Center of Rocket Motor.

experimental data. Again it is noted that, in actuality, the motor cools faster than the predicted value. The maximum difference in temperatures on the surface of the motor is 15°F. Figure 13 shows that the predicted value and the experimental value of the temperature at the center of the rocket motor were in close approximation except during the early stages of the cooling phase where a maximum temperature variation of about 5°F occurred.

One of the reasons the system cools faster than predicted could be the light breeze that is usually evident in the early afternoon hours at China Lake that is not present during the morning. No attempt was made to shield the system from the wind to study the effects of a light breeze on the surface temperature of the storage container.

Another point not taken into account by the analytical model is the fact that the time delay at any point in the system is not constant throughout the day as predicted in Figure 10, but varies as given by the data in Appendix D. Time delays between the peak temperature on the container surface and the peak temperature at the center of the rocket motor vary from about 250 to 400 minutes, whereas the low temperature on the surface of the container and the low temperature at the center of the rocket motor vary from about 150 to 250 minutes. The analytical model predicts a constant variation of 388 minutes at the center of the rocket motor and 159 minutes at the surface.

B. TRUMP MODEL

1. One Dimensional

Four variations of the one dimensional TRUMP model were investigated and compared to the experimental data. Figures 14 and 15 compare the TRUMP predictions to the actual experimental data at the surface and the center of the rocket motor, respectively. The TRUMP variation used for this comparison modeled the storage container temperature with tabular data (See Figure 11) and assumed convection was present ($\frac{k_c}{k} = 4.0$). The uncertainty analysis (Appendix E) established the uncertainty bounds for both the experimental and the analytical data in these Figures. The variation between the bulk temperature predicted by TRUMP and the experimental data closely matches with only two experimental points in Figure 14 falling outside the uncertainty bounds for this one dimensional model. Figure 11 shows that the tabular data that TRUMP interpolates is a good approximation to the averaged experimental data. At the center of the motor, as shown in Figure 15, all experimental points fall within the predicted error bounds. A comparison of the four one-dimensional TRUMP variations are given in Figures 16 and 17 at the surface and the center of the rocket motor respectively. It is clearly seen from these Figures that the convection assumption results in an increase of 2°F in the maximum temperature and a decrease of 2°F in the minimum temperature on the surface of the rocket motor. This temperature change drops to + 1.5°F at the center of the

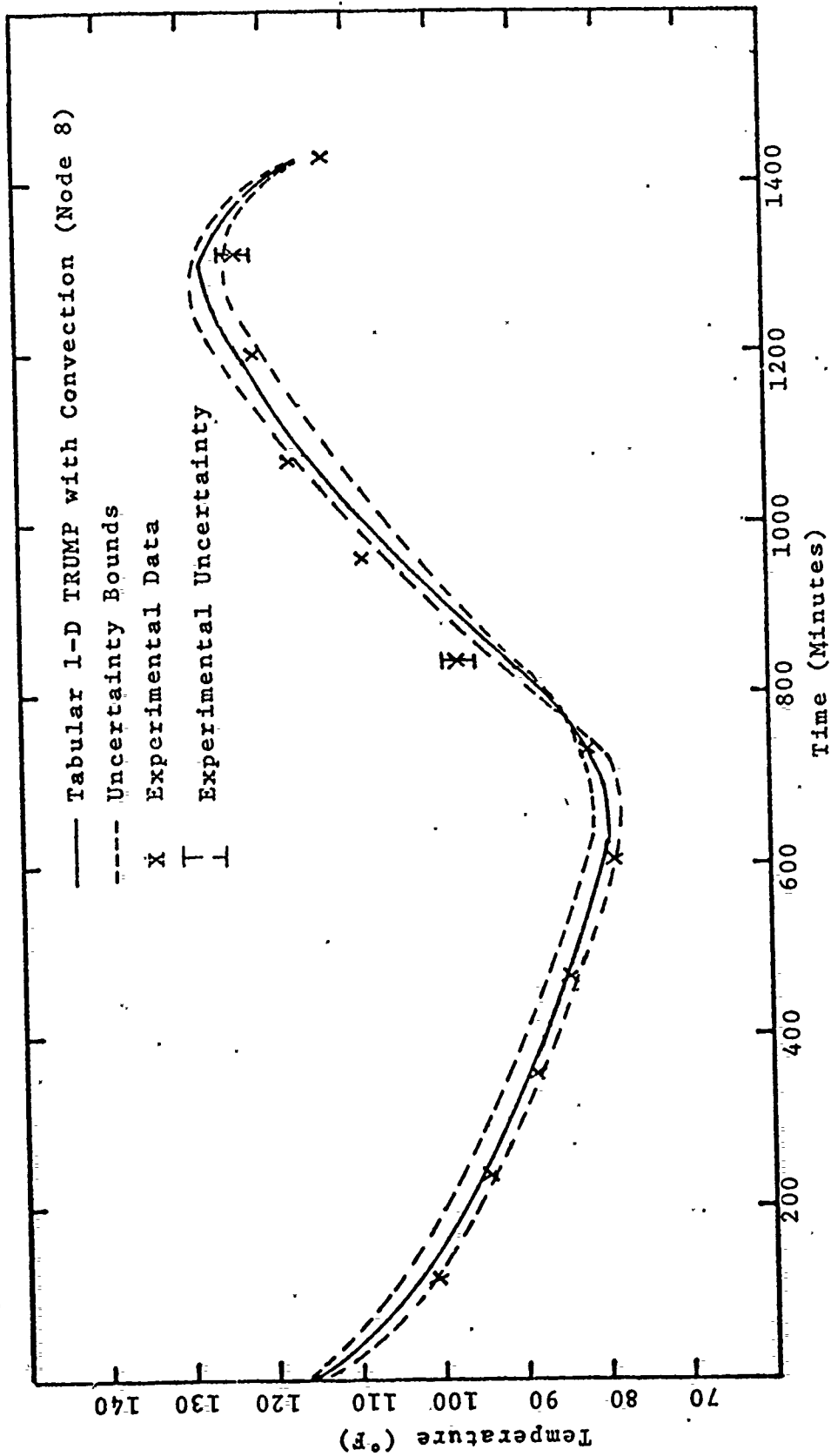


Figure 14: Comparison of 1-D TRUMP and Experimental Temperatures at Surface of the Rocket Motor.

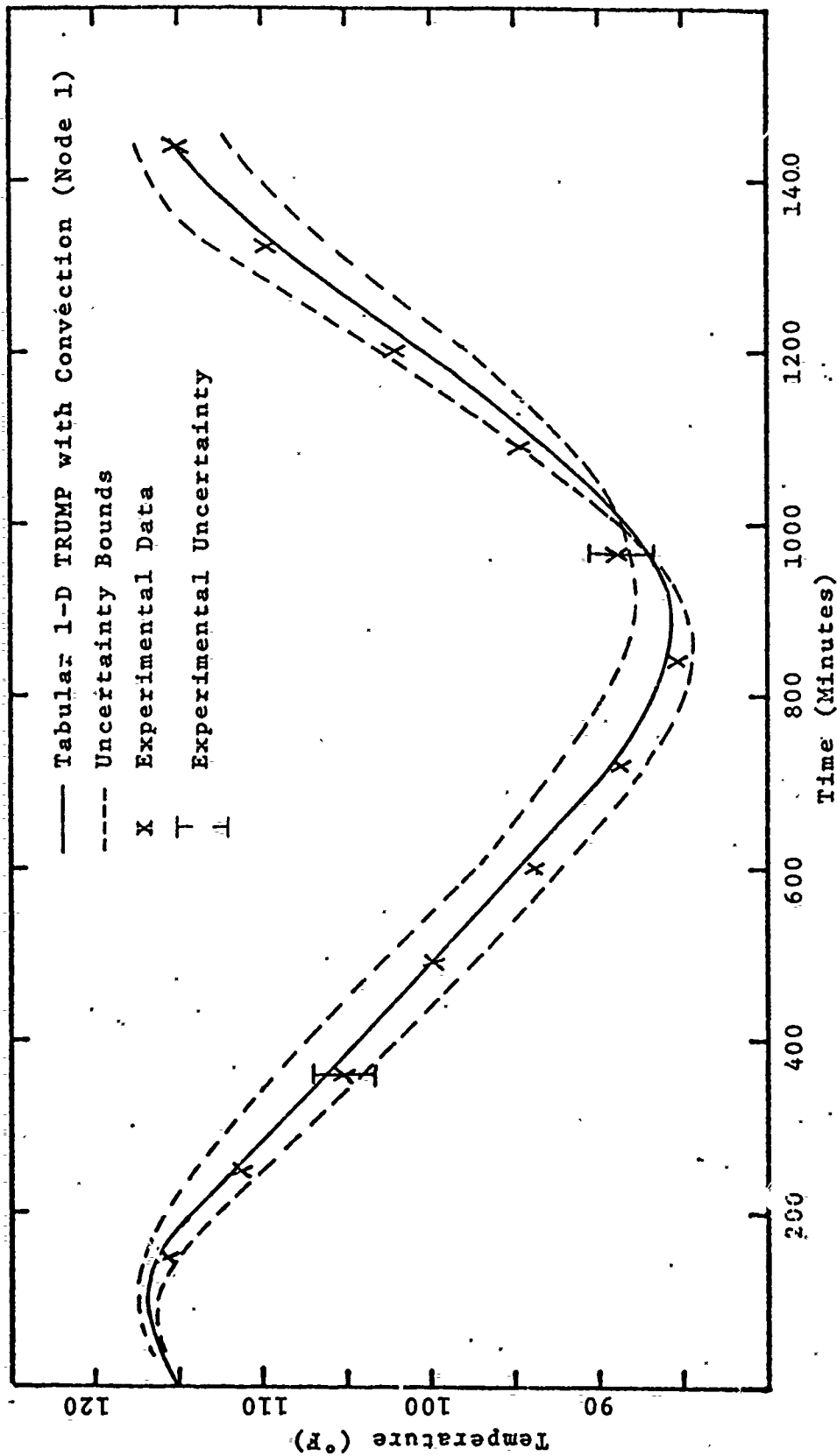


Figure 15: Comparison of 1-D TRUMP and Experimental Temperatures at Center of Rocket Motor.

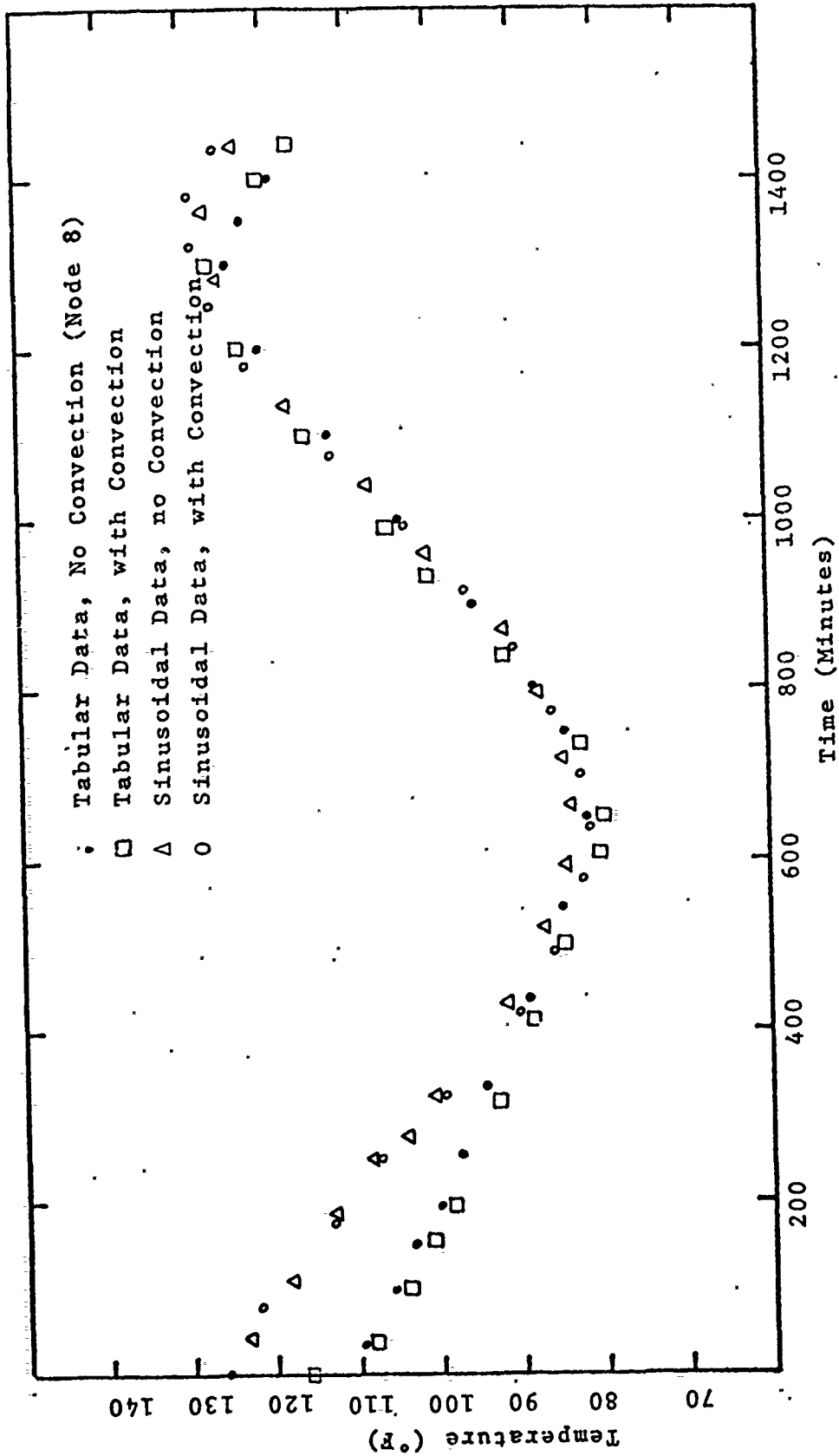


Figure 16: Comparison of Temperatures from Four 1-D TRUMP Variations at Surface of Rocket Motor.

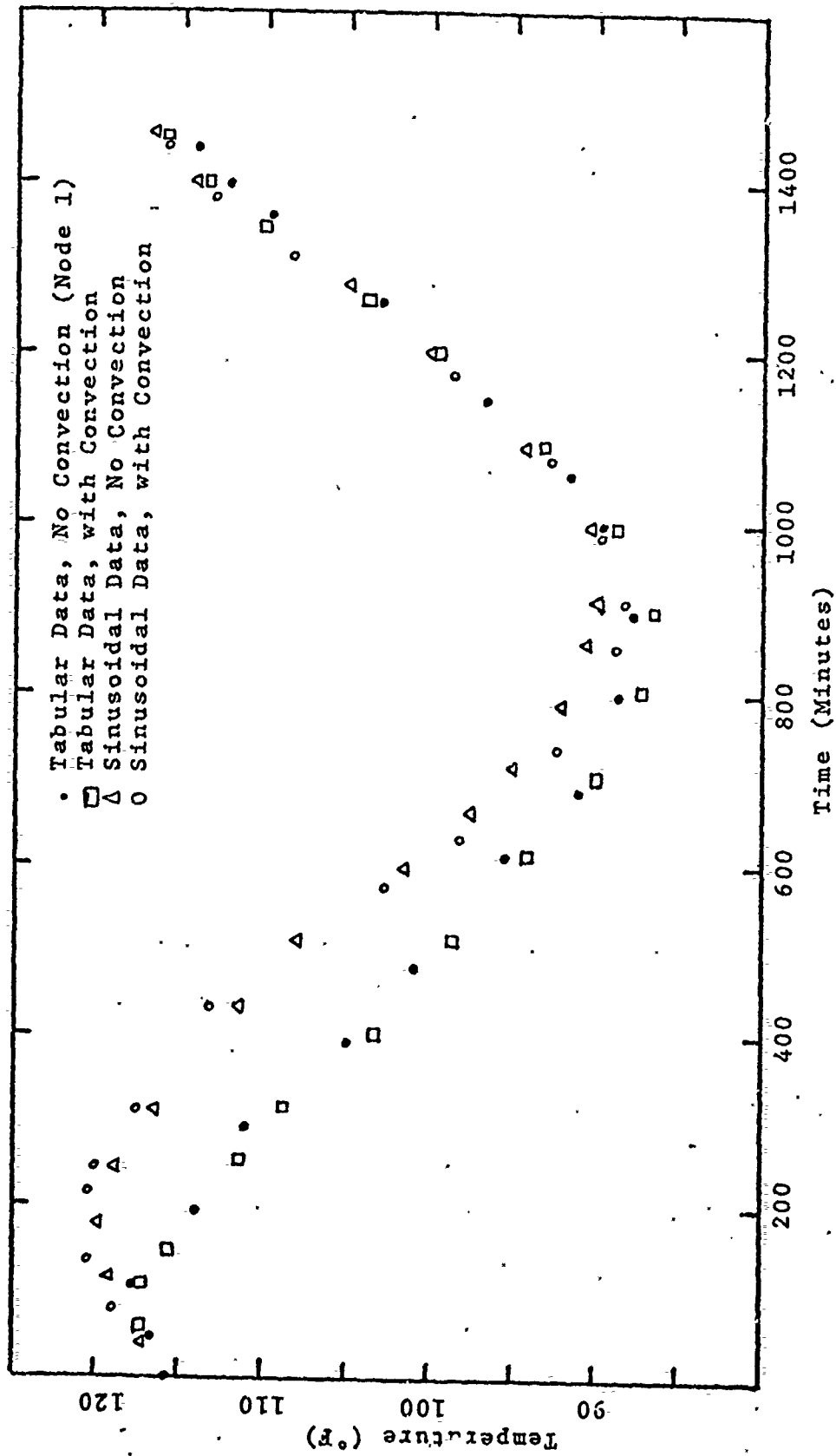


Figure 17: Comparison of Temperatures from Four I-D TRUMP Variations at Center of Rocket Motor.

motor as shown in Figure 17. The differences between the sinusoidal approximation and the tabular approximation of the experimental data was clearly shown in Figure 11. The data in Figures 16 and 17 can be easily correlated to that in Figure 11, thereby explaining the differences in the predicted values.

2. Two Dimensional

Four variations of the two dimensional TRUMP model were investigated and compared to the experimental data. Comparisons of each TRUMP variation to the experimental data are given in Figures 18 and 19 for node 8 (located on the skin of the rocket motor at the 1200 position) and node 1 (at the center of the rocket motor) respectively. These Figures show that the TRUMP variations that used tabular data to model the surface temperature of the storage container predicted temperatures that more closely approximated the experimental values than were those predicted by TRUMP variations using sinusoidal data to model the surface temperature. Appendix D shows that all points on the surface of the storage container reach their minimum temperature at the same time; however, these points reach their maximum temperature as much as 200 minutes apart. Whereas, all the points on the surface of the storage container are in phase at the minimum temperature, they rapidly become out of phase as the container temperature rises. This varying phase shift makes it difficult to model the four boundary nodes with sinusoidal approximations which must have constant

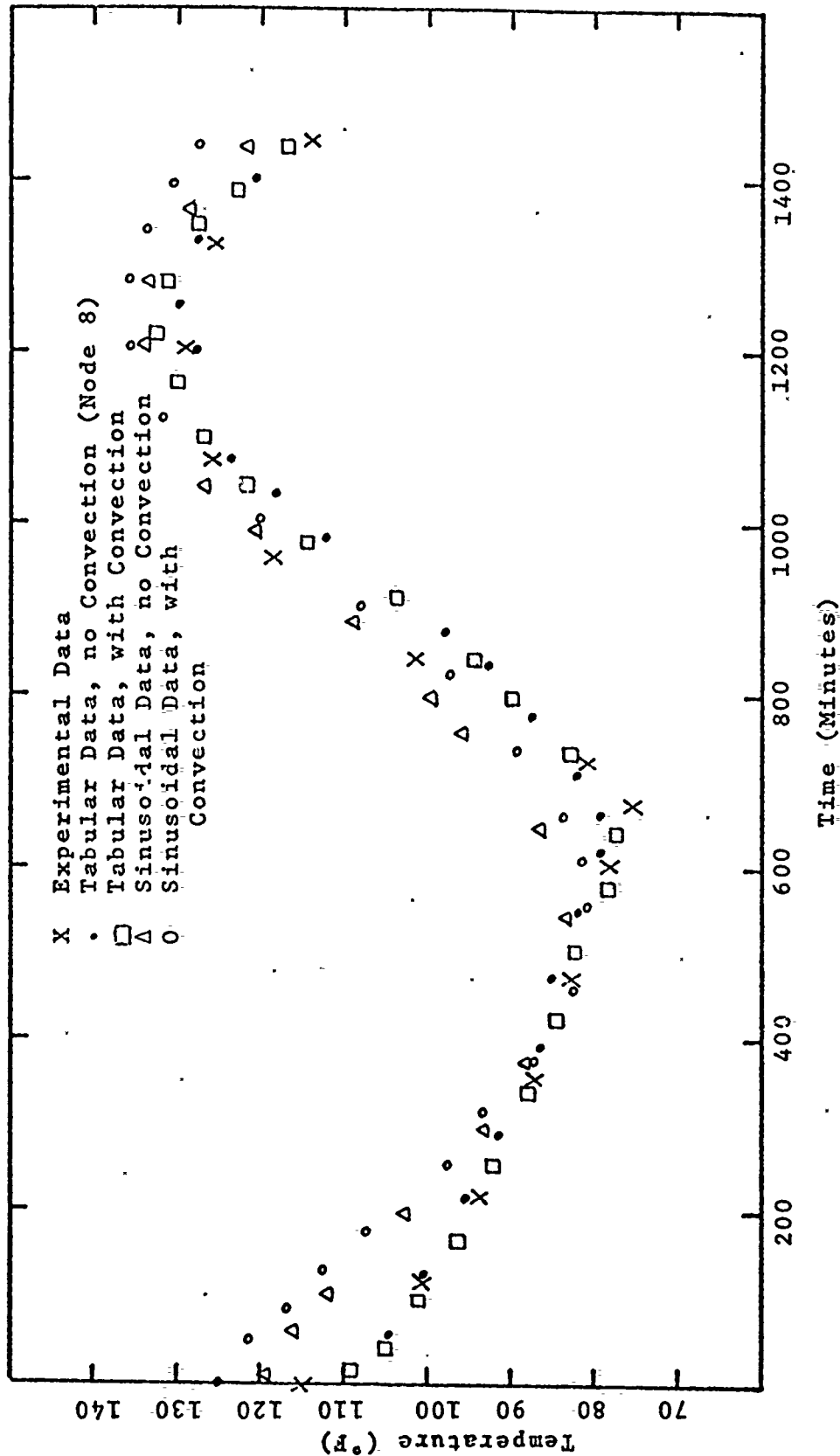


Figure 18: Comparison of 2-D TRUMP and Experimental Temperatures at Surface of Rocket Motor.

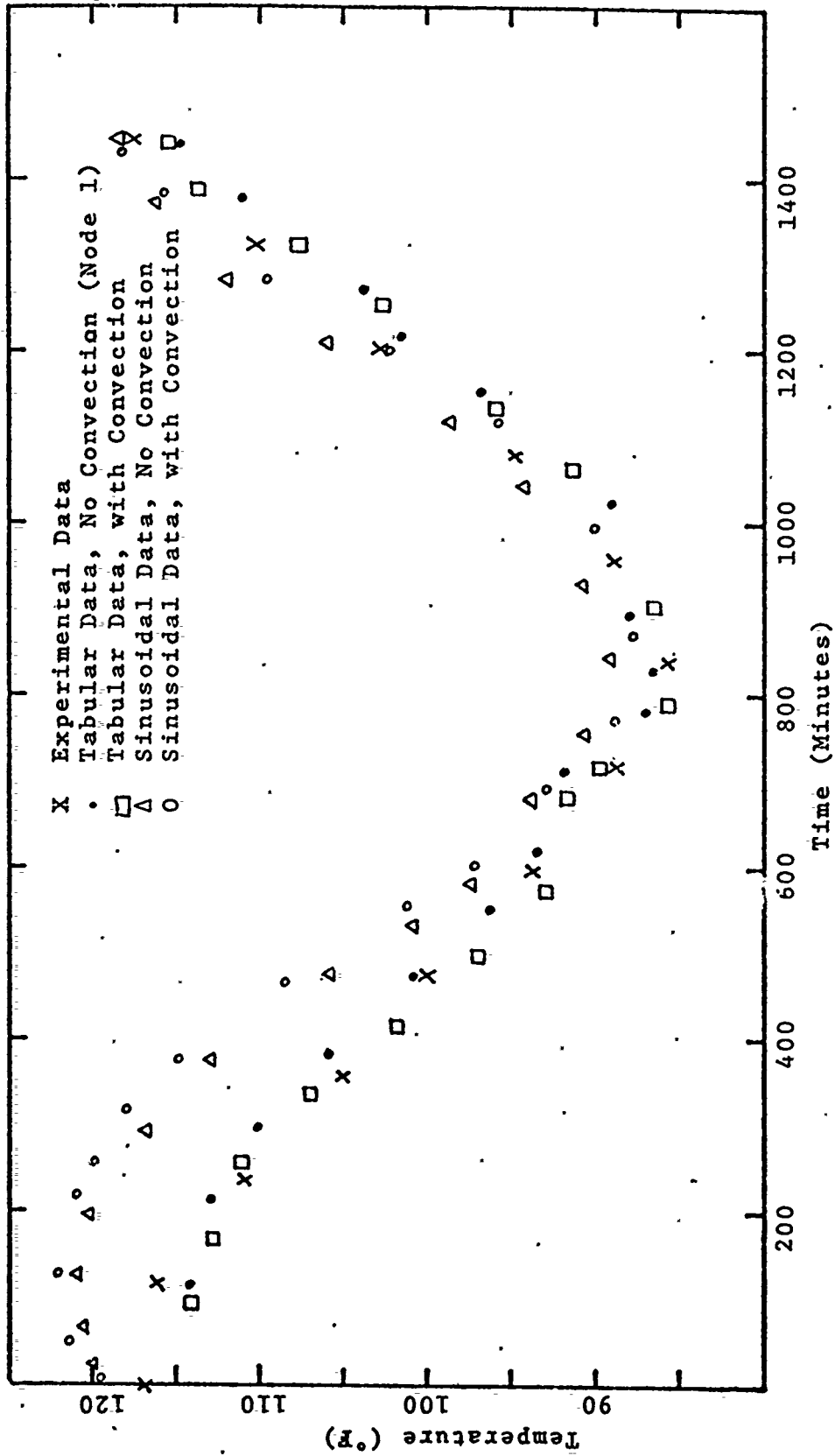


Figure 19: Comparison of 2-D TRUMP and Experimental Temperatures at Center of Rocket Motor.

phase shifts. Sizable errors in the input data during some parts of the cycle were caused by these varying phase shifts. These errors in the input data led to the variations in the predicted temperature values. As noted in the one dimensional section, the inclusion of convection effects does not produce large variations in the predicted temperatures.

Figures 20 and 21 show the actual temperature distributions on the surface of the storage container and on the surface of the rocket motor respectively at maximum bulk temperature compared to a two dimensional TRUMP program. The TRUMP variation used for this comparison assumed no free convection in the air gap and used tabular data to approximate the surface temperature of the storage container.

C. GENERAL

A comparison was made between surface temperatures on the storage container that contained the rocket motor and the storage container that was empty. The low temperature was about 4°F colder in the empty container, whereas the high temperature was about 4°F higher on the container that contained the rocket motor. The empty container had a faster response time than the one containing the motor. The differences in heat capacities, radiation effects, and natural convection all contribute to the changes in temperature noted.

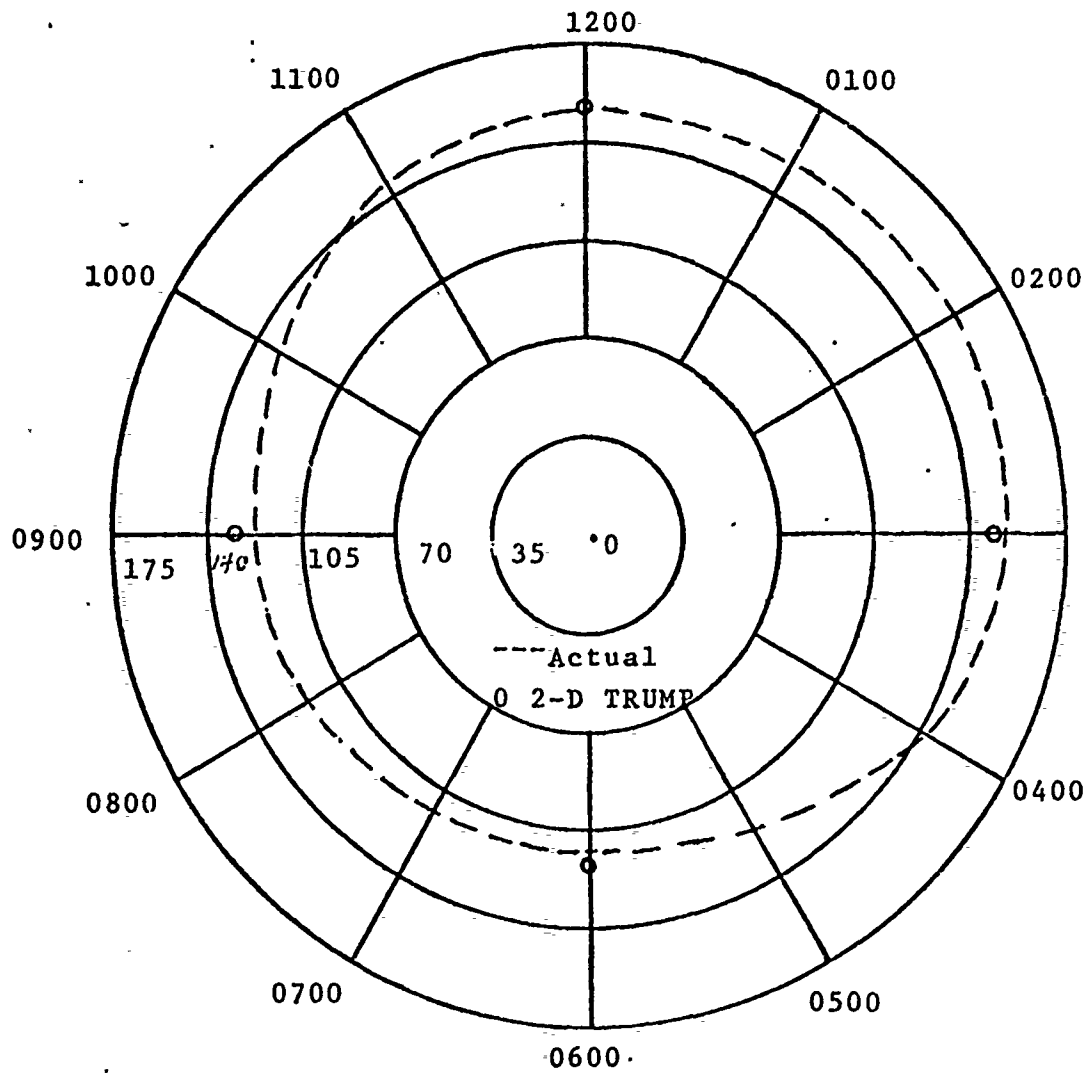


Figure 20: Temperature Distribution at Surface of Storage Container at Maximum Bulk Temperature at approximately 1500 on 2 August 1972.

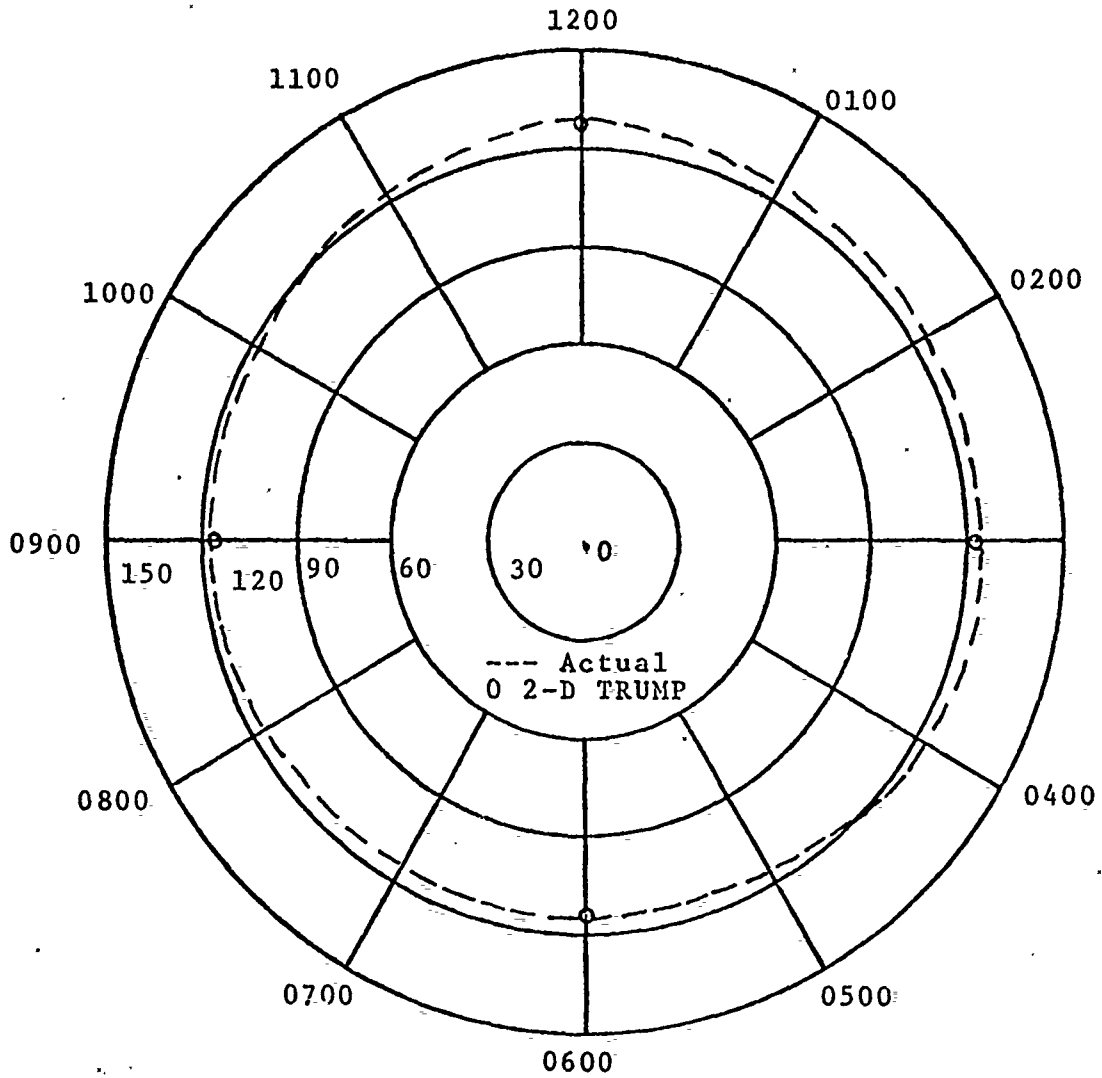


Figure 21: Temperature Distribution at Surface of the Rocket Motor at Maximum Bulk Temperature at approximately 1500 on 2 August 1972.

D. LIQUID CRYSTALS

The encapsulated cholesteric liquid crystals applied to the surface of the storage container gave brilliant colors under the intense desert sun. These colors were much clearer and brighter than the same crystals viewed under laboratory lighting conditions. The liquid crystals photographed well in both the color movies and the color slides. The movies showed by time lapse photography the rapidly changing surface temperature of the storage container. Two sample color prints made from the color slides are enclosed as Figures 22 and 23 to show the brilliance of the colors and the feasibility of obtaining data from color photos. The only photographic problem encountered was the intense reflection of the sunlight from the polyurethane film. This problem was partially overcome by taking the photographs from angles where the reflection was less intense. Qualitatively the liquid crystals were not adversely affected by the sun's rays after two weeks of desert exposure. No accurate quantitative determination was attempted; however, rough approximations were made at the site. These approximations were made by noting the color exhibited by a crystal at a certain time and then comparing the calibration of the crystal (Table 1) to the temperature recorded by the thermocouple located directly beneath the region of color change. The readings were within $\pm 2^{\circ}\text{F}$, which was very encouraging, especially considering the approximations made while taking these measurements. Although photos were taken only during



Figure 22. Thermal Mapping with Liquid Crystals.

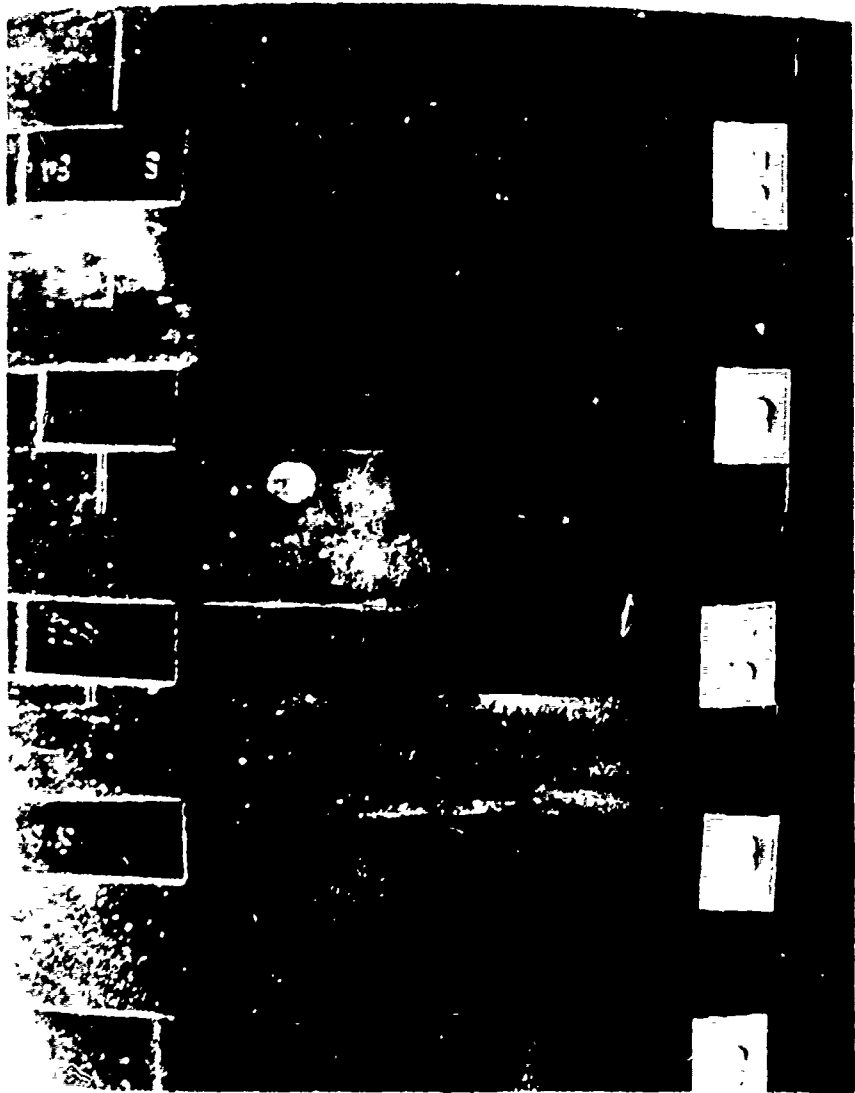


Figure 23. Liquid Crystals Feasible Under Hostile Environment.

the initial two weeks of the study, on site observations indicate that the crystals are still showing brilliant colors after 3-1/2 months. Preliminary evidence indicates that the polyurethane film did protect the crystals from decomposing from the sun's rays and from being worn away by the wind blown sand.

It was noted that the surface temperature of the storage container under the liquid crystals reached temperatures up to 15°F higher than a similar point not under the crystals. This 15°F difference was only evident when maximum temperatures were obtained. During sunlight hours the temperature under the liquid crystals was always somewhat higher; however, at night both temperatures were equal. The difference in the container surface temperatures led to a difference of 4°F on the surface of the rocket motor and 1°F at the center of the motor. It is believed that the difference in emissivities of the gray and black surfaces resulted in the difference in container surface temperature.

VI. CONCLUSIONS

From the results of this investigation, the following conclusions were drawn:

1. Although a sine wave is not a perfect fit for the experimental data at all points, it is useful in predicting bulk temperatures in the rocket motor, especially if only the high and low bulk temperatures are of concern. This is especially true in the one dimensional case. In the two dimensional case, the problem of phase shift variations make the method of sinusoidal variation less desirable, although still useful.

2. The simulation of the actual data by a table of temperatures gave the most accurate predictions of the experimental data. This method should be used whenever tabulated data are available; however, this will generally not be the case for design work, in which case the sinusoidal approximations must be used.

3. The flexibility of both the analytical and computer simulations allow the changing of many parameters. The resulting effects of these changes on rocket motor temperature can be studied with the models.

4. The convection assumption for this system resulted in only small changes in temperature and can be neglected when predicting design temperatures. Either the Liu or Beckmann correlation should be used to determine if convection can be neglected in a particular system.

5. The use of an empty storage container to obtain surface temperature data is a good approximation to using one with a rocket motor inside.

6. It is feasible to use liquid crystals for thermal mapping under desert conditions. Color photography with standard equipment gives excellent results since brilliant colors were observed.

7. The liquid crystals appear to be stable for at least two weeks under the desert conditions when protected with a polyurethane coating.

8. The application of the liquid crystal system to the surface of the storage container resulted in large increases in the surface temperature of the container throughout the hottest part of the day. Care must be taken in applying and interpreting thermal readings from liquid crystals when exposed to radiant heating.

VII. RECOMMENDATIONS

From the results of this basic study, the following recommendations for future work are offered:

1. To refine the results of this project, a second rocket motor storage container system should be instrumented with the following changes:

a. Liquid crystals should not be applied to the system used as the experimental model. As steel is a good thermal conductor, axial conduction on the surface of the storage container may be significant. Heat flow from the area where the crystals are applied may lead to higher than normal temperatures at other points on the surface of the container.

b. The rocket motor should be weighed before and after the loading of the dry sand so that an accurate determination of the density of the propellant simulant can be determined.

c. Four additional thermocouples should be located on the surface of the storage container to better enable the averaging of data. At present, the #1 thermocouple which was used as the average temperature reading of the top quarter of the surface of the container, in actuality is its hottest point; likewise the #10 thermocouple was used as the average temperature of the bottom quarter of the surface, in actuality it's the coldest point. For averaging data, it is recommended that thermocouples be placed at 0130,

0430, 0730, and 1030 and the quarters of the system be divided at 0300, 0600, 0900, and 1200 to give a more realistic bulk temperature. Thermocouples at 1200 and 0600 will provide the maximum and minimum temperature of the system.

2. The TRUMP program should be rerun in both the one and two dimensional form, varying the mesh sizes to determine the optimum number of nodes.

3. A long term study of the effects of the desert environment on liquid crystals should be done. The crystals should be calibrated before being placed in the desert and then brought to a laboratory for recalibration at specific intervals.

4. Several modifications should be made to the TRUMP program to make it comparable to the version used at Lawrence Radiation Laboratory. The variable conductivity section (BLOCK 2) and the PLOT subroutine (BLOCK 11) need to be corrected. The TIMEP subroutine which allows the setting of the problem time interval between data output should be added to this version of TRUMP. It would also be advantageous to increase the amount of tabular data that could be read in as boundary temperatures.

5. From an academic standpoint, the effects of free convection in an air gap with varying boundary temperatures should be investigated.

APPENDIX A

Introduction to Liquid Crystals

Liquid crystals were first discovered in 1889 by Reinitzer [Ref. 8] and the investigations of Lehmann which continued to 1915. Liquid crystals were considered to be laboratory curiosities with no scientific or practical merit until the 1950's. They share some of the properties of both liquids and crystals; for example, a typical liquid crystal substance scatters light in symmetrical patterns and reflects different colors depending on the angle from which it is viewed. Studies in the last few years have helped to clarify the unusual molecular structure of liquid crystals. Many applications arise from their ability to detect minute fluctuations in temperature, mechanical stress, electromagnetic radiation and chemical environment by changes in their color.

Liquid crystals are divided into three classes; smectic, nematic, and cholesteric, depending on the degree of spatial arrangement of the molecules in the mass of the material and the type of the material [Ref. 9]. In this project only cholesteric liquid crystals were used and therefore only their properties will be mentioned. The molecular structure of cholesteric liquid crystals is characteristic of the esters of cholesterol (Figure 24). The molecular layers are very thin with the long axis of the molecules parallel to the plane of the layers. The individual molecules are

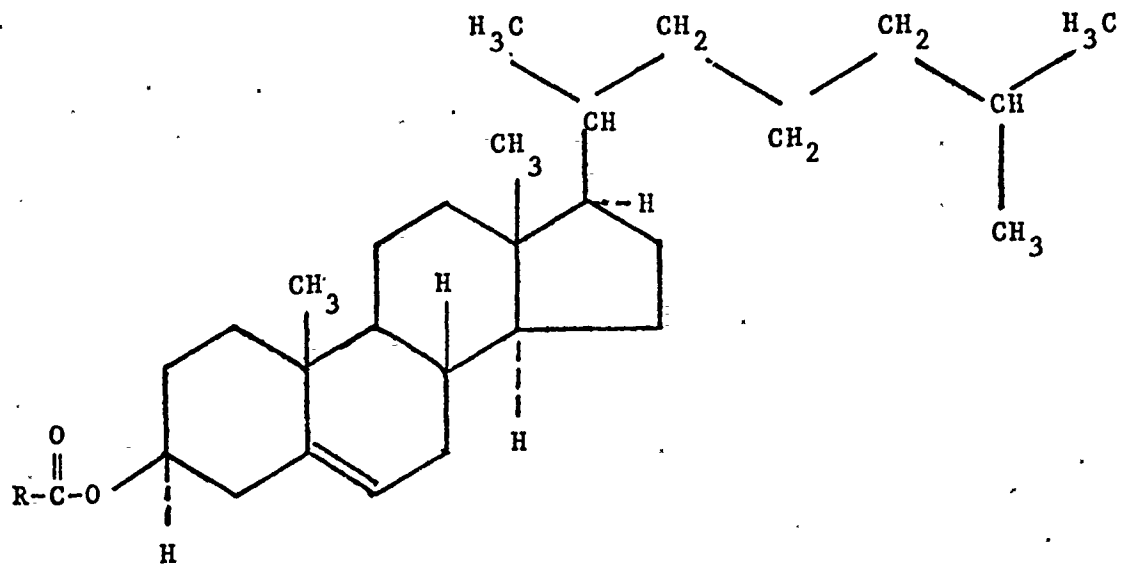


Figure 24: Molecular Structure of Cholesteric Ester

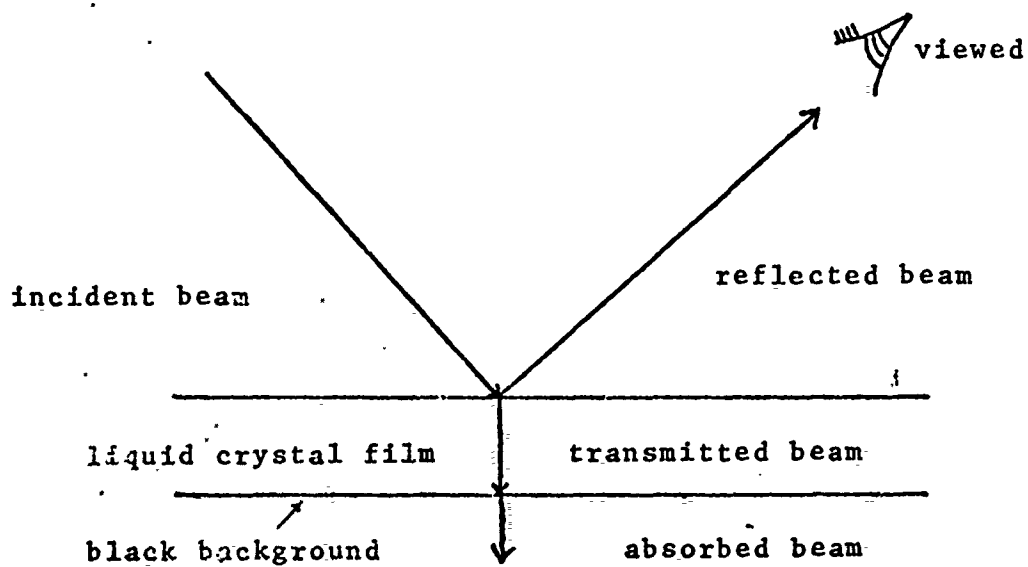


Figure 25. Light Reflection from Liquid Crystals.

basically flat, with a side chain of methyl groups ($-\text{CH}_3$) projecting upward from the plane of each molecule. This configuration causes the direction of the long axis of the molecules in each layer to be displaced slightly from the corresponding direction in adjacent layers. This displacement, which averages about fifteen minutes of arc per layer, is cumulative through successive layers, and the overall displacement traces out a helical path.

The molecular structure of cholesteric liquid crystals gives rise to many peculiar optical properties. If linearly polarized light is transmitted perpendicularly to the molecular layers, the direction of the electric vector of the light will be rotated to the left in a helical path. Therefore, the plane of polarization will also be rotated to the left, through an angle proportional to the thickness of the transmitting material. Liquid crystals are the most optically active substances known. Another strictly crystalline optical property exhibited by cholesteric liquid crystals is circular dichroism. When ordinary white light is incident to a cholesteric material, the light is separated into two components, one with the electric vector rotating clockwise, the other rotating counterclockwise. Depending on the material, one of these components is transmitted, and the other is reflected. It is this property that gives the cholesteric phase its iridescent color when it is illuminated by white light. The particular combination of colors depends on the material, the temperature, and the angle of the incident light.

The molecular structure of a cholesteric substance is very delicately balanced and is easily upset. Any small disturbance that interferes with the weak forces between the molecules can produce marked changes in optical properties such as reflection, transmission, birefringence, circular dichroism, optical activity and color. The most striking optical transformation that occurs in a cholesteric substance, in response to small changes in its environment, is the variation of color with temperature. The crystal lattice is disrupted by the thermal vibrations giving successive transitions between the solid, the mesophase, and the isotropic liquid with rising temperature. The change from the three dimensional order of the crystal lattice to the disorder of the isotropic liquid occurs via one or more intermediate states, each of which has a particular temperature range at which it is stable [Ref. 10].

A cholesteric liquid crystal system responds to changes in temperature by sequentially passing through the complete visual spectrum (red through violet) in fractions or multi-degrees, depending on which cholesterol esters comprise the formulation. This color phenomenon is reversible and has been reported to function over a temperature range of -20°C to 250°C . A very important point to note is that at a certain temperature a given material or combination of materials will always exhibit the same color. Also, the rate of change from color to color as well as the exact temperature at which the specific color changes occur are invariable. By

mixing cholesteric substances in various proportions, any desired temperature combination can be obtained. The thickness of the cholesteric film does not affect the predominant wave length of the reflected light; the light becomes circularly polarized [Ref. 11].

The colors scattered by the liquid crystals represent only a fraction of the incident light (Figure 25). The remaining portion of the incident light is transmitted by the liquid crystals. Therefore, an absorptive black background must be used to prevent reflection of the transmitted light, thereby enhancing the resolution of the scattered colors or wavelengths reflected by the liquid crystal system.

The cholesteric liquid crystal systems often present a number of problems due to the fact that they are viscous liquids. Some problems associated with the handling and the use of these materials are:

1. The tendency of the liquid crystal system to flow during application can cause variations in applied film thickness. This may result in non-uniform thermal patterns.
2. Direct exposure of liquid crystals to adverse environmental effects can cause variations in their sensitivity and deteriorate their color response in a few days.

These problems can be partially overcome by using an encapsulated liquid crystal material system. The capsules are 20-30 microns in diameter and are a water-based slurry suitable for application by conventional coating techniques such as brushing or spraying.

Encapsulated liquid crystals offer several advantages:

1. They convert the liquid crystal system to a pseudo-solid, which provides for easier handling, application, and use.
2. They provide longer shelf life by minimizing surface contamination and giving protection from ultraviolet light [Ref. 12].
3. They exhibit relatively unlimited fatigue life.
4. They reduce the angular dependence of the color observed.

APPENDIX B

Analytical Solution

The method of complex temperature as presented by Arpacı [Ref. 1] was used to find the steady periodic solution of a body experiencing a periodic sinusoidal disturbance. The general heat conduction equation in cylindrical coordinates was the basis for this derivation. It was assumed that no heat sources existed in this problem, that the rocket motor storage container system was infinitely long, that there was no heat conduction in the axial or circumferential directions, and that the container surface temperature was spatially uniform. Figure 26 gives a basic sketch of the system. The assumptions reduced the heat conduction equation to

$$\frac{1}{r} \frac{\partial (r \frac{\partial T}{\partial r})}{\partial r} = \frac{1}{\alpha} \frac{\partial T}{\partial t} \quad (1)$$

where r is the radial distance from the center of the rocket motor, T is the temperature of the rocket motor at time t and position r , and α is the thermal diffusivity, a property of the conducting material.

$$\alpha = \frac{k}{\rho c} \quad (2)$$

where k is the thermal conductivity of the conducting material, ρ is the density of the material, and c is the specific heat. All thermal properties were assumed to be constant over the temperature range of this problem.

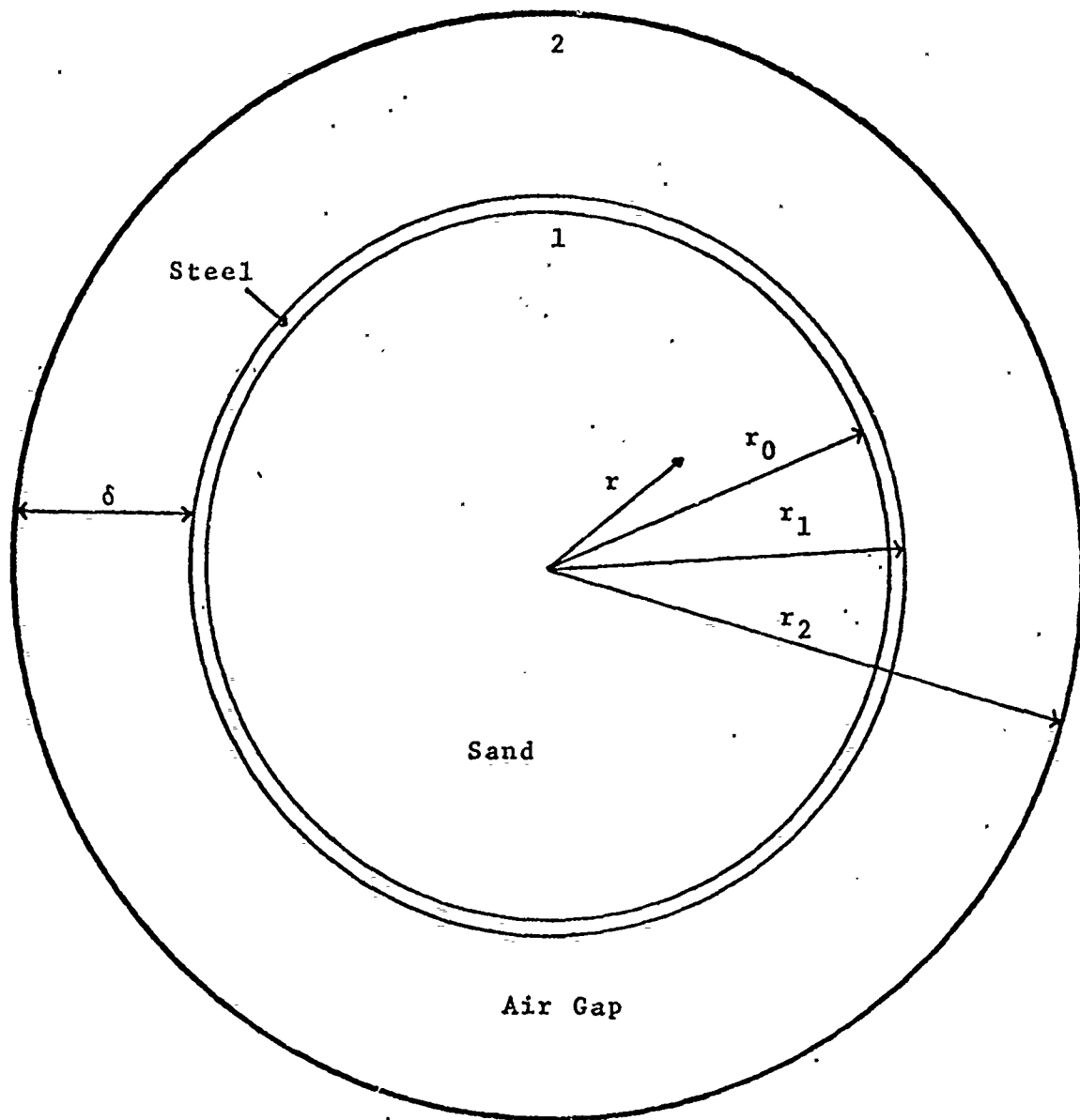


Figure 26. Analytical Model of Experimental System.

The boundary conditions used in this derivation were

$$\frac{dT}{dr} = 0 \quad \text{at } r = 0$$

$$\text{and } \frac{dT}{dr} = -\frac{\bar{h}}{k} (T - T_{\infty}) \quad \text{at } r = r_0$$

where r_0 is the inner radius of the rocket motor.

T_{∞} is the known storage container temperature which is assumed to vary as

$$T_{\infty} = (T_M - T_A) \sin \omega t + T_A$$

where

- T_M = maximum bulk temperature of the storage container
- T_A = average bulk temperature of the storage container
- ω = frequency of the sinusoidal variation ($\frac{2\pi}{24 \text{ hours}}$)
- t = time

\bar{h} is the effective heat transfer coefficient across the air gap between the storage container and the rocket motor. It combines the heat transfer effects of radiation, convection, and conduction into one coefficient. The radiation coefficient was linearized by assuming constant temperatures (T_1, T_2), representative of the average temperatures expected in the problem, in the equation

$$h_{\text{RAD}} = \mathcal{F}_{1-2} \sigma (T_1 + T_2) (T_1^2 + T_2^2)$$

where σ is the Stefan-Boltzmann constant and \mathcal{F}_{1-2} is the radiation exchange factor between surfaces 1 and 2. The convection coefficient is

$$h_{\text{CON}} = \frac{k_c}{\delta}$$

where k_c is the effective conductivity of air as obtained from the Beckmann and Liu correlations [Ref. 5 and 7] and δ is the width of the air gap. In the analytical model,

the effective conductivity was assumed to equal the conductivity, thereby treating it as pure conduction and

$$\bar{h} = h_{\text{RAD}} + h_{\text{CON}}$$

Equation (1) was non-dimensionalized using the following relationships

$$\theta = \frac{T - T_A}{T_M - T_A} \quad (\text{a non-dimensional temperature})$$

$$\xi = \frac{r}{r_o} \quad (\text{a non-dimensional distance})$$

to give

$$\frac{1}{\xi} \frac{d(\xi \frac{d\theta}{d\xi})}{d\xi} = \frac{r_o^2}{\alpha} \frac{d\theta}{dt}$$

with boundary conditions

$$\frac{d\theta}{d\xi} = 0 \quad \text{at } \xi = 0$$

$$\text{and } \frac{d\theta}{d\xi} = -\beta(\theta - \sin \omega t) \text{ at } \xi = 1$$

where $\beta = \frac{\bar{h}r_o}{k}$ is the Biot modulus (which compares the relative magnitudes of the effective heat transfer coefficient across the air gap and the internal conduction resistances to heat transfer).

An initial condition was not specified as the only concern was with the steady state, periodic behavior. Following Arpaci [Ref. 1], a complex temperature was defined as

$$\psi(r,t) = \theta^*(r,t) + i\theta(r,t)$$

where $\psi(r,t)$ satisfied

$$\frac{1}{\xi} \frac{d(\xi \frac{d\psi}{d\xi})}{d\xi} = \frac{r_o^2}{\alpha} \frac{d\psi}{dt} \quad (3)$$

with boundary conditions

$$\frac{d\psi}{d\xi} = \frac{d\theta^*}{d\xi} + i \frac{d\theta}{d\xi} = 0 \quad \text{at } \xi = 0$$

$$\text{and } \frac{d\psi}{d\xi} = -\beta(\psi - e^{i\omega t}) = -\beta(\theta^* - \cos \omega t) + i\{-\beta(\theta - \sin \omega t)\} \text{ at } \xi = 1$$

This leads to $\theta(r, t)$ which satisfied

$$\frac{1}{\xi} \frac{d(\xi \frac{d\theta}{d\xi})}{d\xi} = \frac{r_0^2}{\alpha} \frac{d\theta}{dt}$$

with boundary conditions

$$\frac{d\theta}{d\xi} = 0 \quad \text{at } \xi = 0$$

$$\text{and } \frac{d\theta}{d\xi} = -\beta(\theta - \sin \omega t) \quad \text{at } \xi = 1$$

also $\theta^*(r, t)$ which satisfied

$$\frac{1}{\xi} \frac{d(\xi \frac{d\theta^*}{d\xi})}{d\xi} = \frac{r_0^2}{\alpha} \frac{d\theta^*}{dt}$$

with boundary conditions

$$\frac{d\theta^*}{d\xi} = 0 \quad \text{at } \xi = 0$$

$$\text{and } \frac{d\theta^*}{d\xi} = -\beta(\theta^* - \cos \omega t) \quad \text{at } \xi = 1$$

A solution of the form

$$\psi(r, t) = \phi(r)\tau(t)$$

was assumed, where for large values of time $\tau(t)$ was assumed to equal $e^{i\omega t}$; therefore,

$$\psi(r, t) = \phi(r)e^{i\omega t} \quad (4)$$

Equation (4) was then substituted into equation (3)

$$\frac{1}{\xi} \frac{d(\xi \frac{d\phi}{d\xi})}{d\xi} - \frac{i\omega r_0^2 \phi}{\alpha} = 0 \quad (5)$$

with boundary conditions

$$\frac{d\phi}{d\xi} = 0 \quad \text{at } \xi = 0$$

and
$$\frac{d\phi}{d\xi} = -\beta(\phi-1) \quad \text{at } \xi = 1$$

Equation (5) was expanded to give

$$\frac{d^2\phi}{d\xi^2} + \frac{1}{\xi} \frac{d\phi}{d\xi} - \frac{i\omega r_o^2}{\alpha} \phi = 0 \quad (6)$$

Now, let
$$Z = \sqrt{\frac{i\omega r_o^2}{\alpha}} \xi$$

and substitute into equation (6)

$$\frac{d^2\phi}{dZ^2} + \frac{1}{Z} \frac{d\phi}{dZ} - \phi = 0 \quad (7)$$

with boundary conditions

$$\frac{d\phi}{dZ} = 0 \quad \text{at } Z = 0$$

and

$$\frac{d\phi}{dZ} = -\frac{\beta}{\sqrt{\frac{i\omega r_o^2}{\alpha}}} (\phi-1) \quad \text{at } Z = \sqrt{\frac{i\omega r_o^2}{\alpha}}$$

The general solution of equation (7) is

$$\phi = C_1 I_0(Z) + C_2 K_0(Z) \quad (8)$$

as given in Ref. 13 with

$$I_0(Z) = 1 + \left(\frac{1}{2}Z\right)^2 + \frac{\left(\frac{1}{2}Z\right)^4}{(2!)^2} + \dots = \sum_{n=0}^{\infty} \frac{\left(\frac{1}{2}Z\right)^{2n}}{(n!)^2}$$

and

$$K_0(Z) = -\{\gamma + \log\left(\frac{1}{2}Z\right)\} I_0(Z) + \sum_{n=1}^{\infty} \frac{\left(\frac{1}{2}Z\right)^{2n}}{(n!)^2} \left\{1 + \frac{1}{2} + \frac{1}{3} + \dots + \frac{1}{n}\right\}$$

Now using the boundary condition

$$\frac{d\phi}{dZ} = 0 \quad \text{at } Z = 0$$

and differentiating equation (8) yields

$$\frac{d\phi}{dz} = c_1 \frac{d(I_o(z))}{dz} + c_2 \frac{d(K_o(z))}{dz}$$

where

$$\frac{d(I_o(z))}{dz} = 0 \quad \text{at } z = 0$$

and

$$\frac{d(K_o(z))}{dz} \neq 0 \quad \text{at } z = 0$$

therefore $c_2 \equiv 0$

$$\text{and } \phi = c_1 I_o(z) \quad (9)$$

Now using the second boundary condition that

$$\frac{d\phi}{dz} = - \frac{\beta}{\sqrt{\frac{i\omega r_o^2}{\alpha}}} (\phi - 1) \quad \text{at } z = \sqrt{\frac{i\omega r_o^2}{\alpha}} \quad (10)$$

and differentiating equation (9) gives

$$\frac{d\phi}{dz} = c_1 \frac{d(I_o(z))}{dz}$$

Noting that $\frac{d(I_o(z))}{dz} = I_1(z)$ and substituting into equation (10)

$$c_1 I_1 \left(\sqrt{\frac{i\omega r_o^2}{\alpha}} \right) = \frac{\beta}{\sqrt{\frac{i\omega r_o^2}{\alpha}}} \left(c_1 I_o \left(\sqrt{\frac{i\omega r_o^2}{\alpha}} \right) - 1 \right)$$

Rearranging and solving for c_1

$$c_1 = \frac{1}{\sqrt{\frac{i\omega r_o^2}{\alpha\beta^2}} I_1 \left(\sqrt{\frac{i\omega r_o^2}{\alpha}} \right) + I_o \left(\sqrt{\frac{i\omega r_o^2}{\alpha}} \right)}$$

and then substituting into equation (9)

$$\phi = \frac{I_0(z)}{I_0\left(\sqrt{\frac{i\omega r_o^2}{\alpha}}\right) + \frac{1}{\beta}\sqrt{\frac{i\omega r_o^2}{\alpha}} I_1\left(\sqrt{\frac{i\omega r_o^2}{\alpha}}\right)} \quad (11)$$

Now as

$$J_\nu(imx) = i^\nu I_\nu(mx) \quad [\text{Ref. 3, p. 135}]$$

$$I_0\left(\sqrt{\frac{\omega r_o^2}{\alpha}} i^{1/2} \xi\right) = J_0\left(\sqrt{\frac{\omega r_o^2}{\alpha}} i^{3/2} \xi\right)$$

and

$$I_1\left(\sqrt{\frac{\omega r_o^2}{\alpha}} i^{1/2} \xi\right) = \frac{1}{i} J_1\left(\sqrt{\frac{\omega r_o^2}{\alpha}} i^{3/2} \xi\right)$$

Let $a = \sqrt{\frac{\omega r_o^2}{\alpha}}$ and substitute into equation (11)

$$\phi = \frac{J_0(i^{3/2} a\xi)}{J_0(i^{3/2} a) - \frac{a}{\beta} i^{3/2} J_1(i^{3/2} a)} \quad (12)$$

$$\text{Now } i^{3/2} = e^{i\frac{3\pi}{4}} = \cos \frac{3\pi}{4} + i \sin \frac{3\pi}{4} = \frac{1}{\sqrt{2}} (-1 + i)$$

Substituting this into equation (12)

$$\phi = \frac{J_0(i^{3/2} a\xi)}{J_0(i^{3/2} a) + \frac{a}{\sqrt{2}\beta} (1-i) J_1(i^{3/2} a)} \quad (13)$$

$$\text{As } J_0(a\xi i^{3/2}) = J_0(a\xi e^{i\frac{3\pi}{4}}) = \text{BER}_0(a\xi) + i\text{BEI}_0(a\xi)$$

and

$$J_1(a\xi i^{3/2}) = J_1(a\xi e^{i\frac{3\pi}{4}}) = \text{BER}_1(a\xi) + i\text{BEI}_1(a\xi)$$

Substituting these results into equation (13) yields

$$\phi = \frac{\text{BER}_0(a\xi) + i\text{BEI}_0(a\xi)}{\text{BER}_0(a) + i\text{BEI}_0(a) + \frac{a}{\sqrt{2}\beta} (1-i) (\text{BER}_1(a) + i\text{BEI}_1(a))} \quad (14)$$

After rearrangement.

$$\phi = \frac{\text{BER}_0(a\xi) + i\text{BEi}_0(a\xi)}{[\text{BER}_0(a) + \frac{a}{\sqrt{2\beta}}\text{BER}_1(a) + \frac{a}{\sqrt{2\beta}}\text{BEi}_1(a)] + i[\text{BEi}_0(a) + \frac{a}{\sqrt{2\beta}}\text{BEi}_1(a) - \frac{a}{\sqrt{2\beta}}\text{BER}_1(a)]}$$

Letting

$$X_R = \text{BER}_0(a) + \frac{a}{\sqrt{2\beta}}\text{BER}_1(a) + \frac{a}{\sqrt{2\beta}}\text{BEi}_1(a)$$

and

$$X_i = \text{BEi}_0(a) + \frac{a}{\sqrt{2\beta}}\text{BEi}_1(a) - \frac{a}{\sqrt{2\beta}}\text{BER}_1(a)$$

and substituting into equation (14) gives

$$\phi = \frac{\text{BER}_0(a\xi) + i\text{BEi}_0(a\xi)}{X_R + iX_i}$$

Rationalizing the denominator yields

$$\phi = \frac{\text{BER}_0(a\xi) + i\text{BEi}_0(a\xi)}{X_R^2 + X_i^2} (X_R - iX_i) \quad (15)$$

Now

$$\phi = \frac{(\text{BER}_0(a\xi)X_R + \text{BEi}_0(a\xi)X_i) + i(\text{BEi}_0(a\xi)X_R - \text{BER}_0(a\xi)X_i)}{X_R^2 + X_i^2}$$

which after rearrangement gives

$$\phi = \sqrt{\frac{\text{BER}_0^2(a\xi) + \text{BEi}_0^2(a\xi)}{X_R^2 + X_i^2}} e^{i\delta^*} \quad (16)$$

where

$$\delta^* = \tan^{-1} \frac{\text{BEi}_0(a\xi)X_R - \text{BER}_0(a\xi)X_i}{\text{BER}_0(a\xi)X_R + \text{BEi}_0(a\xi)X_i}$$

Substituting into equation (4) gives

$$\psi(r,t) = \sqrt{\frac{BER_o^2(a\xi) + BEi_o^2(a\xi)}{X_R^2 + X_i^2}} e^{i(\omega t + \delta^*)}$$

which also equals

$$\psi(r,t) = \sqrt{\frac{BER_o^2(a\xi) + BEi_o^2(a\xi)}{X_R^2 + X_i^2}} [\cos(\omega t + \delta^*) + i \sin(\omega t + \delta^*)]$$

As this problem was modeled as a sine wave, the imaginary part of $\psi(r,t)$ was used.

$$I(\psi(r,t)) = \theta(r,t) = \sqrt{\frac{BER_o^2(a\xi) + BEi_o^2(a\xi)}{X_R^2 + X_i^2}} \sin(\omega t + \delta^*) \quad (17)$$

which is the analytical solution of infinitely long concentric cylinders experiencing a periodic sinusoidal temperature variation on its outermost surface when heat conduction is assumed to be radial only.

In summary

$$\theta(r,t) = \frac{T - T_A}{T_M - T_A} = \sqrt{\frac{BER_o^2(a\xi) + BEi_o^2(a\xi)}{X_R^2 + X_i^2}} \sin(\omega t + \delta^*)$$

where

T = the temperature of a point r in the rocket motor at time t

T_A = average bulk temperature of the storage container

T_M = maximum bulk temperature of the storage container

ω = frequency of the sinusoidal variation ($2\pi/24$ hours)

t = time

$\xi = \frac{r}{r_o}$ = dimensionless distance from the center of the rocket motor

r_o = distance to the surface of the rocket motor

r = distance from the center of the rocket motor

$a = \sqrt{\frac{\omega r_o^2}{\alpha}}$ = conduction parameter

$$\alpha = \frac{k}{\rho c} = \text{thermal diffusivity}$$

ρ = density

k = thermal conductivity

c = specific heat

BER = real Bessel Function

BEi = imaginary Bessel Function

$$X_R = \text{BER}_0(a) + \frac{a}{\sqrt{2\beta}} \text{BER}_1(a) + \frac{a}{\sqrt{2\beta}} \text{BEi}_1(a)$$

$$X_i = \frac{\text{BEi}_0(a)}{hr_0} + \frac{a}{\sqrt{2\beta}} \text{BEi}_1(a) - \frac{a}{\sqrt{2\beta}} \text{BER}_1(a)$$

$$\beta = \frac{hr_0}{k} = \text{Biot modulus}$$

$$\delta^* = \tan^{-1} \frac{\text{BEi}_0(a\xi)X_R - \text{BER}_0(a\xi)X_i}{\text{BER}_0(a\xi)X_R + \text{BEi}_0(a\xi)X_i}$$

The following computer programs were used to investigate a wide variety of parameters. The outputs are samples of some of these parameter studies.

ANALYTICAL SOLUTION PARAMETER STUDY

```

0 2 K=1.5
READ(5,11)A
FORMAT(F10.6)
11 WRITE(6,31)
31 FORMAT(1,19X,'A',4X,'B',4X,'DISTANCE FROM CENTER',4X,'TIME DELAY
1 IN RADIANS',3X,'RELATIVE AMPLITUDE')
DO 3 L=1,10
READ(5,11)B
FROM INPUT VALUES OF A AND B, CALCULATE THE DENOMINATOR(DENOM) OF EQUATION 17.
Y=SQRT(2.0)
Z=A
BEROZ=(1.0-(Z**4/64.0)+(Z**8/147500.0))
BERIOZ=0.25*Z**2*(1.0-(Z**4/576.0)+(Z**8/3790000.0))
BERIIZ=-Z/(2.0*Y)*(1.0+(Z**2/8.0)-(Z**6/9216.0)+(Z**8/
1737280.0))
1 BEIIZ=Z/(2.0*Y)*(1.0-(Z**2/8.0)-(Z**4/128.0)+(Z**6/9216.0)+(Z**8/7
137280.0))
XR=BEROZ+A/(Y*8)*BERIIZ+A/(Y*8)*BERIIZ
XI=BERIOZ+A/(Y*8)*BERIIZ-A/(Y*8)*BERIIZ
V=XR**2+XI**2
DENOM=SQRT(V)
AT THE CENTER, HALF WAY TO THE SURFACE, AND AT THE SURFACE OF THE MOTOR,
CALCULATE THE NUMERATOR(DNUM) OF EQUATION 17.
DO 30 I=1,3
C=(I-1)/2.0
Z=A*C
BEROZ=(1.0-(Z**4/64.0)+(Z**8/147500.0))
BERIOZ=0.25*Z**2*(1.0-(Z**4/576.0)+(Z**8/3790000.0))
U=BEROZ**2+BERIOZ**2
DNUM=SQRT(U)
CALCULATE THE TIME DELAY(DEL).
DNU=XR*BERIOZ-XI*BEROZ
DNO=XR*BEROZ+XI*BERIIZ
X=DNU/DNO
IF(DNO.LT.0.0)GO TO 22
DEL=A*ATAN(X)
GO TO 23
22 DEL=ATAN(X)-3.14159
CALCULATE THE RELATIVE AMPLITUDE(S).
23 S=DNUM/DENOM
WRITE(6,32)A,B,C,DEL,S
32 FORMAT(1,19X,F3.1,1X,F5.1,12X,F3.1,19X,F5.2,18X,F4.2)
30 CONTINUE
3 CONTINUE
2 STOP
END

```


| A | B | DISTANCE FROM CENTER | TIME DELAY IN RADIAN | RELATIVE AMPLITUDE |
|---|---|----------------------|----------------------|--------------------|
| 2 | 0 | 0.5 | 2.78 | 0.05 |
| 2 | 0 | 1.0 | 1.12 | 0.06 |
| 2 | 0 | 1.5 | 1.59 | 0.22 |
| 2 | 0 | 2.0 | 1.92 | 0.27 |
| 2 | 1 | 0.5 | 1.64 | 0.37 |
| 2 | 1 | 1.0 | 1.74 | 0.45 |
| 2 | 1 | 1.5 | 1.43 | 0.54 |
| 2 | 2 | 0.5 | 1.52 | 0.66 |
| 2 | 2 | 1.0 | 1.31 | 0.65 |
| 2 | 2 | 1.5 | 1.09 | 0.76 |
| 2 | 3 | 0.5 | 1.23 | 0.67 |
| 2 | 3 | 1.0 | 1.38 | 0.82 |
| 2 | 3 | 1.5 | 1.73 | 0.70 |
| 2 | 4 | 0.5 | 1.19 | 0.71 |
| 2 | 4 | 1.0 | 1.25 | 0.86 |
| 2 | 4 | 1.5 | 1.81 | 0.77 |
| 2 | 5 | 0.5 | 1.14 | 0.93 |
| 2 | 5 | 1.0 | 1.49 | 0.80 |
| 2 | 5 | 1.5 | 1.63 | 0.82 |
| 2 | 6 | 0.5 | 1.09 | 0.89 |
| 2 | 6 | 1.0 | 1.60 | 0.81 |
| 2 | 6 | 1.5 | 1.61 | 0.99 |

| A | B | DISTANCE FROM CENTER | TIME DELAY IN RADIAN | RELATIVE AMPLITUDE |
|-----|-------|----------------------|----------------------|--------------------|
| 4.0 | 0.0 | 0.5 | 3.64 | 0.01 |
| 4.0 | 0.1 | 1.0 | 2.74 | 0.01 |
| 4.0 | 0.15 | 1.0 | 2.25 | 0.03 |
| 4.0 | 0.5 | 1.0 | 1.56 | 0.04 |
| 4.0 | 1.0 | 1.0 | 1.15 | 0.10 |
| 4.0 | 1.5 | 1.0 | 0.95 | 0.16 |
| 4.0 | 2.0 | 1.0 | 0.82 | 0.22 |
| 4.0 | 2.5 | 1.0 | 0.73 | 0.27 |
| 4.0 | 3.0 | 1.0 | 0.68 | 0.31 |
| 4.0 | 3.5 | 1.0 | 0.65 | 0.35 |
| 4.0 | 4.0 | 1.0 | 0.63 | 0.38 |
| 4.0 | 4.5 | 1.0 | 0.62 | 0.41 |
| 4.0 | 5.0 | 1.0 | 0.61 | 0.43 |
| 4.0 | 5.5 | 1.0 | 0.61 | 0.45 |
| 4.0 | 6.0 | 1.0 | 0.61 | 0.47 |
| 4.0 | 6.5 | 1.0 | 0.61 | 0.49 |
| 4.0 | 7.0 | 1.0 | 0.61 | 0.51 |
| 4.0 | 7.5 | 1.0 | 0.61 | 0.53 |
| 4.0 | 8.0 | 1.0 | 0.61 | 0.55 |
| 4.0 | 8.5 | 1.0 | 0.61 | 0.57 |
| 4.0 | 9.0 | 1.0 | 0.61 | 0.59 |
| 4.0 | 9.5 | 1.0 | 0.61 | 0.61 |
| 4.0 | 10.0 | 1.0 | 0.61 | 0.63 |
| 4.0 | 10.5 | 1.0 | 0.61 | 0.65 |
| 4.0 | 11.0 | 1.0 | 0.61 | 0.67 |
| 4.0 | 11.5 | 1.0 | 0.61 | 0.69 |
| 4.0 | 12.0 | 1.0 | 0.61 | 0.71 |
| 4.0 | 12.5 | 1.0 | 0.61 | 0.73 |
| 4.0 | 13.0 | 1.0 | 0.61 | 0.75 |
| 4.0 | 13.5 | 1.0 | 0.61 | 0.77 |
| 4.0 | 14.0 | 1.0 | 0.61 | 0.79 |
| 4.0 | 14.5 | 1.0 | 0.61 | 0.81 |
| 4.0 | 15.0 | 1.0 | 0.61 | 0.83 |
| 4.0 | 15.5 | 1.0 | 0.61 | 0.85 |
| 4.0 | 16.0 | 1.0 | 0.61 | 0.87 |
| 4.0 | 16.5 | 1.0 | 0.61 | 0.89 |
| 4.0 | 17.0 | 1.0 | 0.61 | 0.91 |
| 4.0 | 17.5 | 1.0 | 0.61 | 0.93 |
| 4.0 | 18.0 | 1.0 | 0.61 | 0.95 |
| 4.0 | 18.5 | 1.0 | 0.61 | 0.97 |
| 4.0 | 19.0 | 1.0 | 0.61 | 0.99 |
| 4.0 | 19.5 | 1.0 | 0.61 | 1.01 |
| 4.0 | 20.0 | 1.0 | 0.61 | 1.03 |
| 4.0 | 20.5 | 1.0 | 0.61 | 1.05 |
| 4.0 | 21.0 | 1.0 | 0.61 | 1.07 |
| 4.0 | 21.5 | 1.0 | 0.61 | 1.09 |
| 4.0 | 22.0 | 1.0 | 0.61 | 1.11 |
| 4.0 | 22.5 | 1.0 | 0.61 | 1.13 |
| 4.0 | 23.0 | 1.0 | 0.61 | 1.15 |
| 4.0 | 23.5 | 1.0 | 0.61 | 1.17 |
| 4.0 | 24.0 | 1.0 | 0.61 | 1.19 |
| 4.0 | 24.5 | 1.0 | 0.61 | 1.21 |
| 4.0 | 25.0 | 1.0 | 0.61 | 1.23 |
| 4.0 | 25.5 | 1.0 | 0.61 | 1.25 |
| 4.0 | 26.0 | 1.0 | 0.61 | 1.27 |
| 4.0 | 26.5 | 1.0 | 0.61 | 1.29 |
| 4.0 | 27.0 | 1.0 | 0.61 | 1.31 |
| 4.0 | 27.5 | 1.0 | 0.61 | 1.33 |
| 4.0 | 28.0 | 1.0 | 0.61 | 1.35 |
| 4.0 | 28.5 | 1.0 | 0.61 | 1.37 |
| 4.0 | 29.0 | 1.0 | 0.61 | 1.39 |
| 4.0 | 29.5 | 1.0 | 0.61 | 1.41 |
| 4.0 | 30.0 | 1.0 | 0.61 | 1.43 |
| 4.0 | 30.5 | 1.0 | 0.61 | 1.45 |
| 4.0 | 31.0 | 1.0 | 0.61 | 1.47 |
| 4.0 | 31.5 | 1.0 | 0.61 | 1.49 |
| 4.0 | 32.0 | 1.0 | 0.61 | 1.51 |
| 4.0 | 32.5 | 1.0 | 0.61 | 1.53 |
| 4.0 | 33.0 | 1.0 | 0.61 | 1.55 |
| 4.0 | 33.5 | 1.0 | 0.61 | 1.57 |
| 4.0 | 34.0 | 1.0 | 0.61 | 1.59 |
| 4.0 | 34.5 | 1.0 | 0.61 | 1.61 |
| 4.0 | 35.0 | 1.0 | 0.61 | 1.63 |
| 4.0 | 35.5 | 1.0 | 0.61 | 1.65 |
| 4.0 | 36.0 | 1.0 | 0.61 | 1.67 |
| 4.0 | 36.5 | 1.0 | 0.61 | 1.69 |
| 4.0 | 37.0 | 1.0 | 0.61 | 1.71 |
| 4.0 | 37.5 | 1.0 | 0.61 | 1.73 |
| 4.0 | 38.0 | 1.0 | 0.61 | 1.75 |
| 4.0 | 38.5 | 1.0 | 0.61 | 1.77 |
| 4.0 | 39.0 | 1.0 | 0.61 | 1.79 |
| 4.0 | 39.5 | 1.0 | 0.61 | 1.81 |
| 4.0 | 40.0 | 1.0 | 0.61 | 1.83 |
| 4.0 | 40.5 | 1.0 | 0.61 | 1.85 |
| 4.0 | 41.0 | 1.0 | 0.61 | 1.87 |
| 4.0 | 41.5 | 1.0 | 0.61 | 1.89 |
| 4.0 | 42.0 | 1.0 | 0.61 | 1.91 |
| 4.0 | 42.5 | 1.0 | 0.61 | 1.93 |
| 4.0 | 43.0 | 1.0 | 0.61 | 1.95 |
| 4.0 | 43.5 | 1.0 | 0.61 | 1.97 |
| 4.0 | 44.0 | 1.0 | 0.61 | 1.99 |
| 4.0 | 44.5 | 1.0 | 0.61 | 2.01 |
| 4.0 | 45.0 | 1.0 | 0.61 | 2.03 |
| 4.0 | 45.5 | 1.0 | 0.61 | 2.05 |
| 4.0 | 46.0 | 1.0 | 0.61 | 2.07 |
| 4.0 | 46.5 | 1.0 | 0.61 | 2.09 |
| 4.0 | 47.0 | 1.0 | 0.61 | 2.11 |
| 4.0 | 47.5 | 1.0 | 0.61 | 2.13 |
| 4.0 | 48.0 | 1.0 | 0.61 | 2.15 |
| 4.0 | 48.5 | 1.0 | 0.61 | 2.17 |
| 4.0 | 49.0 | 1.0 | 0.61 | 2.19 |
| 4.0 | 49.5 | 1.0 | 0.61 | 2.21 |
| 4.0 | 50.0 | 1.0 | 0.61 | 2.23 |
| 4.0 | 50.5 | 1.0 | 0.61 | 2.25 |
| 4.0 | 51.0 | 1.0 | 0.61 | 2.27 |
| 4.0 | 51.5 | 1.0 | 0.61 | 2.29 |
| 4.0 | 52.0 | 1.0 | 0.61 | 2.31 |
| 4.0 | 52.5 | 1.0 | 0.61 | 2.33 |
| 4.0 | 53.0 | 1.0 | 0.61 | 2.35 |
| 4.0 | 53.5 | 1.0 | 0.61 | 2.37 |
| 4.0 | 54.0 | 1.0 | 0.61 | 2.39 |
| 4.0 | 54.5 | 1.0 | 0.61 | 2.41 |
| 4.0 | 55.0 | 1.0 | 0.61 | 2.43 |
| 4.0 | 55.5 | 1.0 | 0.61 | 2.45 |
| 4.0 | 56.0 | 1.0 | 0.61 | 2.47 |
| 4.0 | 56.5 | 1.0 | 0.61 | 2.49 |
| 4.0 | 57.0 | 1.0 | 0.61 | 2.51 |
| 4.0 | 57.5 | 1.0 | 0.61 | 2.53 |
| 4.0 | 58.0 | 1.0 | 0.61 | 2.55 |
| 4.0 | 58.5 | 1.0 | 0.61 | 2.57 |
| 4.0 | 59.0 | 1.0 | 0.61 | 2.59 |
| 4.0 | 59.5 | 1.0 | 0.61 | 2.61 |
| 4.0 | 60.0 | 1.0 | 0.61 | 2.63 |
| 4.0 | 60.5 | 1.0 | 0.61 | 2.65 |
| 4.0 | 61.0 | 1.0 | 0.61 | 2.67 |
| 4.0 | 61.5 | 1.0 | 0.61 | 2.69 |
| 4.0 | 62.0 | 1.0 | 0.61 | 2.71 |
| 4.0 | 62.5 | 1.0 | 0.61 | 2.73 |
| 4.0 | 63.0 | 1.0 | 0.61 | 2.75 |
| 4.0 | 63.5 | 1.0 | 0.61 | 2.77 |
| 4.0 | 64.0 | 1.0 | 0.61 | 2.79 |
| 4.0 | 64.5 | 1.0 | 0.61 | 2.81 |
| 4.0 | 65.0 | 1.0 | 0.61 | 2.83 |
| 4.0 | 65.5 | 1.0 | 0.61 | 2.85 |
| 4.0 | 66.0 | 1.0 | 0.61 | 2.87 |
| 4.0 | 66.5 | 1.0 | 0.61 | 2.89 |
| 4.0 | 67.0 | 1.0 | 0.61 | 2.91 |
| 4.0 | 67.5 | 1.0 | 0.61 | 2.93 |
| 4.0 | 68.0 | 1.0 | 0.61 | 2.95 |
| 4.0 | 68.5 | 1.0 | 0.61 | 2.97 |
| 4.0 | 69.0 | 1.0 | 0.61 | 2.99 |
| 4.0 | 69.5 | 1.0 | 0.61 | 3.01 |
| 4.0 | 70.0 | 1.0 | 0.61 | 3.03 |
| 4.0 | 70.5 | 1.0 | 0.61 | 3.05 |
| 4.0 | 71.0 | 1.0 | 0.61 | 3.07 |
| 4.0 | 71.5 | 1.0 | 0.61 | 3.09 |
| 4.0 | 72.0 | 1.0 | 0.61 | 3.11 |
| 4.0 | 72.5 | 1.0 | 0.61 | 3.13 |
| 4.0 | 73.0 | 1.0 | 0.61 | 3.15 |
| 4.0 | 73.5 | 1.0 | 0.61 | 3.17 |
| 4.0 | 74.0 | 1.0 | 0.61 | 3.19 |
| 4.0 | 74.5 | 1.0 | 0.61 | 3.21 |
| 4.0 | 75.0 | 1.0 | 0.61 | 3.23 |
| 4.0 | 75.5 | 1.0 | 0.61 | 3.25 |
| 4.0 | 76.0 | 1.0 | 0.61 | 3.27 |
| 4.0 | 76.5 | 1.0 | 0.61 | 3.29 |
| 4.0 | 77.0 | 1.0 | 0.61 | 3.31 |
| 4.0 | 77.5 | 1.0 | 0.61 | 3.33 |
| 4.0 | 78.0 | 1.0 | 0.61 | 3.35 |
| 4.0 | 78.5 | 1.0 | 0.61 | 3.37 |
| 4.0 | 79.0 | 1.0 | 0.61 | 3.39 |
| 4.0 | 79.5 | 1.0 | 0.61 | 3.41 |
| 4.0 | 80.0 | 1.0 | 0.61 | 3.43 |
| 4.0 | 80.5 | 1.0 | 0.61 | 3.45 |
| 4.0 | 81.0 | 1.0 | 0.61 | 3.47 |
| 4.0 | 81.5 | 1.0 | 0.61 | 3.49 |
| 4.0 | 82.0 | 1.0 | 0.61 | 3.51 |
| 4.0 | 82.5 | 1.0 | 0.61 | 3.53 |
| 4.0 | 83.0 | 1.0 | 0.61 | 3.55 |
| 4.0 | 83.5 | 1.0 | 0.61 | 3.57 |
| 4.0 | 84.0 | 1.0 | 0.61 | 3.59 |
| 4.0 | 84.5 | 1.0 | 0.61 | 3.61 |
| 4.0 | 85.0 | 1.0 | 0.61 | 3.63 |
| 4.0 | 85.5 | 1.0 | 0.61 | 3.65 |
| 4.0 | 86.0 | 1.0 | 0.61 | 3.67 |
| 4.0 | 86.5 | 1.0 | 0.61 | 3.69 |
| 4.0 | 87.0 | 1.0 | 0.61 | 3.71 |
| 4.0 | 87.5 | 1.0 | 0.61 | 3.73 |
| 4.0 | 88.0 | 1.0 | 0.61 | 3.75 |
| 4.0 | 88.5 | 1.0 | 0.61 | 3.77 |
| 4.0 | 89.0 | 1.0 | 0.61 | 3.79 |
| 4.0 | 89.5 | 1.0 | 0.61 | 3.81 |
| 4.0 | 90.0 | 1.0 | 0.61 | 3.83 |
| 4.0 | 90.5 | 1.0 | 0.61 | 3.85 |
| 4.0 | 91.0 | 1.0 | 0.61 | 3.87 |
| 4.0 | 91.5 | 1.0 | 0.61 | 3.89 |
| 4.0 | 92.0 | 1.0 | 0.61 | 3.91 |
| 4.0 | 92.5 | 1.0 | 0.61 | 3.93 |
| 4.0 | 93.0 | 1.0 | 0.61 | 3.95 |
| 4.0 | 93.5 | 1.0 | 0.61 | 3.97 |
| 4.0 | 94.0 | 1.0 | 0.61 | 3.99 |
| 4.0 | 94.5 | 1.0 | 0.61 | 4.01 |
| 4.0 | 95.0 | 1.0 | 0.61 | 4.03 |
| 4.0 | 95.5 | 1.0 | 0.61 | 4.05 |
| 4.0 | 96.0 | 1.0 | 0.61 | 4.07 |
| 4.0 | 96.5 | 1.0 | 0.61 | 4.09 |
| 4.0 | 97.0 | 1.0 | 0.61 | 4.11 |
| 4.0 | 97.5 | 1.0 | 0.61 | 4.13 |
| 4.0 | 98.0 | 1.0 | 0.61 | 4.15 |
| 4.0 | 98.5 | 1.0 | 0.61 | 4.17 |
| 4.0 | 99.0 | 1.0 | 0.61 | 4.19 |
| 4.0 | 99.5 | 1.0 | 0.61 | 4.21 |
| 4.0 | 100.0 | 1.0 | 0.61 | 4.23 |

ANALYTICAL SOLUTION + SAMPLE PROBLEM

```

//WIR11687 JOB (1687,0860ET,NF12), 'WIRZBURGER, ALLEN'
// EXEC FOR TCLGP, REGION=150K
// FORT SYS IN DD *
REAL LABEL
REAL#8 ITITLE(I2)
REAL#4 J2
VALUES ARE A, B, MAXIMUM TEMPERATURE AND AVERAGE TEMPERATURE.
DIMENSION J2(49), TINF(49), TI(49), T2(49)
READ(5,11) A, B, TM, TA
FORMAT(4F10.6) (ITITLE(I), I=1, 12)
READ(5,10) (I=1, 12)
FORMAT(6A8)
DMEGA=2.0*3.14159/1440.0
FROM INPUT VALUES OF A AND B, CALCULATE THE DENOMINATOR(DENOM) OF EQUATION 17.
Z=A
BEROZ=(1.0-(Z**4/64.0)+(Z**8/147500.0))
BEIOZ=0.25*Z**2*(1.0-(Z**4/576.0)+(Z**8/3790000.0))
BERIZ=-Z/(2.0*Y)*(1.0+(Z**2/8.0)-(Z**4/128.0)-(Z**6/9216.0)+(Z**8/
1737280.0))
BERIIZ=Z/(2.0*Y)*(1.0-(Z**2/8.0)-(Z**4/128.0)+(Z**6/9216.0)+(Z**8/7
137280.0))
XR=BEROZ+A/(Y*B)*BERIZ+A/(Y*B)*BEIIZ
XI=BEIOZ+A/(Y*B)*BERIZ-A/(Y*B)*BERIZ
V=XR**2+XI**2
DENOM=SQRT(V)
N=0
THE TEMPERATURE OF THE CONTAINER(TINF) IS CALCULATED AT 30 MINUTE INTERVALS.
DO 40 J=1,1441,30
N=N+1
J2(N)=J-1
TEMP=DMEGA*J2(N)
TINF(N)=(TM-TA)*SIN(TEMP)+TA
IF(J.EQ.1441) GO TO 31
L=(N-1)/6
N2=N-1
L2=6*L
IF(N2.EQ.L2) GO TO 21
FOR SEVEN POSITIONS BETWEEN THE CENTER OF THE ROCKET MOTOR AND THE SURFACE, THE
NUMERATOR(DNUM) OF EQUATION 17 AND THE TIME DELAY(DEL) ARE CALCULATED.
DO 30 I=1,7
C=(I-1)/5.75
IF(C.GT.1.0) C=1.0
Z=A*C
BEROZ=(1.0-(Z**4/64.0)+(Z**8/147500.0))
BEIOZ=0.25*Z**2*(1.0-(Z**4/576.0)+(Z**8/3790000.0))

```


| DISTANCE FROM CENTER | TIME | TEMPERATURE | TEMPERATURE OF CONTAINER | TIME DELAY |
|----------------------|-------|-------------|--------------------------|------------|
| 0.00 | 0.00 | 88.28 | 104.00 | 363.40 |
| 1.00 | 0.00 | 88.20 | 104.00 | 378.17 |
| 2.00 | 0.00 | 88.06 | 104.00 | 347.89 |
| 3.00 | 0.00 | 88.19 | 104.00 | 232.10 |
| 4.00 | 0.00 | 88.02 | 104.00 | 159.29 |
| 5.00 | 0.00 | 88.16 | 104.00 | 102.42 |
| 0.00 | 3.00 | 88.17 | 108.44 | 378.17 |
| 1.00 | 3.00 | 88.34 | 108.44 | 347.89 |
| 2.00 | 3.00 | 88.25 | 108.44 | 297.61 |
| 3.00 | 3.00 | 88.45 | 108.44 | 232.10 |
| 4.00 | 3.00 | 88.27 | 108.44 | 159.29 |
| 5.00 | 3.00 | 88.11 | 108.44 | 102.40 |
| 0.00 | 6.00 | 88.34 | 108.80 | 378.17 |
| 1.00 | 6.00 | 88.81 | 108.80 | 347.89 |
| 2.00 | 6.00 | 88.84 | 108.80 | 297.61 |
| 3.00 | 6.00 | 88.57 | 108.80 | 232.10 |
| 4.00 | 6.00 | 88.73 | 108.80 | 159.29 |
| 5.00 | 6.00 | 88.60 | 108.80 | 102.40 |
| 0.00 | 9.00 | 88.91 | 107.00 | 378.17 |
| 1.00 | 9.00 | 88.63 | 107.00 | 347.89 |
| 2.00 | 9.00 | 88.68 | 107.00 | 297.61 |
| 3.00 | 9.00 | 88.97 | 107.00 | 232.10 |
| 4.00 | 9.00 | 89.76 | 107.00 | 159.29 |
| 5.00 | 9.00 | 89.49 | 107.00 | 102.40 |
| 0.00 | 12.00 | 89.62 | 107.00 | 378.17 |
| 1.00 | 12.00 | 89.46 | 107.00 | 347.89 |
| 2.00 | 12.00 | 89.60 | 107.00 | 297.61 |
| 3.00 | 12.00 | 89.56 | 107.00 | 232.10 |
| 4.00 | 12.00 | 89.79 | 107.00 | 159.29 |
| 5.00 | 12.00 | 89.34 | 107.00 | 102.40 |
| 0.00 | 15.00 | 90.37 | 124.70 | 378.17 |
| 1.00 | 15.00 | 91.89 | 124.70 | 347.89 |
| 2.00 | 15.00 | 91.09 | 124.70 | 297.61 |
| 3.00 | 15.00 | 91.72 | 124.70 | 232.10 |
| 4.00 | 15.00 | 91.09 | 124.70 | 159.29 |
| 5.00 | 15.00 | 91.72 | 124.70 | 102.40 |
| 0.00 | 18.00 | 103.77 | 124.70 | 378.17 |
| 1.00 | 18.00 | 108.77 | 124.70 | 347.89 |
| 2.00 | 18.00 | 108.77 | 124.70 | 297.61 |
| 3.00 | 18.00 | 108.77 | 124.70 | 232.10 |
| 4.00 | 18.00 | 108.77 | 124.70 | 159.29 |
| 5.00 | 18.00 | 108.77 | 124.70 | 102.40 |

| DISTANCE FROM CENTER | TIME | TEMPERATURE | TEMPERATURE OF CONTAINER | TIME DELAY |
|----------------------|------|-------------|--------------------------|------------|
| 0 | 180 | 91.50 | 128.04 | 388.47 |
| 1 | 180 | 91.34 | 128.04 | 378.16 |
| 2 | 180 | 95.94 | 128.04 | 47.81 |
| 3 | 180 | 99.85 | 128.04 | 232.10 |
| 4 | 180 | 105.71 | 128.04 | 102.42 |
| 5 | 180 | 111.88 | 128.04 | 388.17 |
| 0 | 210 | 92.38 | 130.97 | 47.81 |
| 1 | 210 | 94.98 | 130.97 | 47.81 |
| 2 | 210 | 97.27 | 130.97 | 232.10 |
| 3 | 210 | 102.42 | 130.97 | 102.42 |
| 4 | 210 | 108.42 | 130.97 | 388.17 |
| 5 | 210 | 114.44 | 133.44 | 47.81 |
| 0 | 240 | 95.77 | 133.44 | 47.81 |
| 1 | 240 | 96.86 | 133.44 | 232.10 |
| 2 | 240 | 104.01 | 133.44 | 102.42 |
| 3 | 240 | 111.10 | 133.44 | 388.17 |
| 4 | 240 | 117.18 | 135.41 | 47.81 |
| 5 | 240 | 123.70 | 135.41 | 47.81 |
| 0 | 270 | 98.07 | 135.41 | 232.10 |
| 1 | 270 | 102.87 | 135.41 | 102.42 |
| 2 | 270 | 106.48 | 135.41 | 388.17 |
| 3 | 270 | 112.48 | 135.41 | 47.81 |
| 4 | 270 | 118.70 | 135.41 | 47.81 |
| 5 | 270 | 125.00 | 135.41 | 232.10 |
| 0 | 300 | 100.71 | 136.84 | 102.42 |
| 1 | 300 | 104.15 | 136.84 | 388.17 |
| 2 | 300 | 109.70 | 136.84 | 47.81 |
| 3 | 300 | 115.60 | 136.84 | 47.81 |
| 4 | 300 | 121.01 | 136.84 | 232.10 |
| 5 | 300 | 127.77 | 136.84 | 102.42 |
| 0 | 330 | 100.69 | 137.71 | 388.17 |
| 1 | 330 | 102.31 | 137.71 | 47.81 |
| 2 | 330 | 106.31 | 137.71 | 232.10 |
| 3 | 330 | 111.74 | 137.71 | 102.42 |
| 4 | 330 | 117.42 | 137.71 | 388.17 |
| 5 | 330 | 123.00 | 137.71 | 47.81 |

| DISTANCE FROM CENTER | TIME | TEMPERATURE | TEMPERATURE DE CONIAINER | TIME DELAY |
|----------------------|------|-------------|--------------------------|------------|
| 0.00 | 00 | 102.04 | 138.00 | 388.40 |
| 1.00 | 60 | 104.86 | 138.00 | 378.17 |
| 2.00 | 00 | 108.41 | 138.00 | 347.61 |
| 3.00 | 60 | 113.59 | 138.00 | 297.89 |
| 4.00 | 00 | 124.90 | 138.00 | 255.10 |
| 5.00 | 60 | 104.18 | 137.71 | 102.40 |
| 0.00 | 90 | 106.94 | 137.71 | 378.17 |
| 1.00 | 00 | 110.45 | 137.71 | 347.61 |
| 2.00 | 60 | 115.25 | 137.71 | 297.89 |
| 3.00 | 00 | 121.03 | 137.71 | 255.10 |
| 4.00 | 60 | 106.18 | 136.84 | 102.40 |
| 5.00 | 00 | 108.57 | 136.84 | 378.17 |
| 0.00 | 30 | 112.94 | 136.84 | 347.61 |
| 1.00 | 60 | 126.78 | 136.84 | 297.89 |
| 2.00 | 00 | 108.21 | 136.84 | 255.10 |
| 3.00 | 60 | 114.89 | 135.41 | 102.40 |
| 4.00 | 00 | 123.74 | 135.41 | 378.17 |
| 5.00 | 60 | 110.87 | 135.41 | 347.61 |
| 0.00 | 90 | 127.16 | 135.41 | 297.89 |
| 1.00 | 00 | 112.59 | 135.41 | 255.10 |
| 2.00 | 60 | 123.99 | 133.44 | 102.40 |
| 3.00 | 00 | 115.37 | 133.44 | 378.17 |
| 4.00 | 60 | 122.63 | 133.44 | 347.61 |
| 5.00 | 00 | 111.27 | 133.44 | 297.89 |
| 0.00 | 30 | 125.69 | 133.44 | 255.10 |
| 1.00 | 60 | 102.63 | 133.44 | 102.40 |
| 2.00 | 00 | 114.99 | 133.44 | 378.17 |
| 3.00 | 60 | 122.47 | 133.44 | 347.61 |
| 4.00 | 00 | 112.47 | 133.44 | 297.89 |
| 5.00 | 60 | 120.26 | 133.44 | 255.10 |
| 0.00 | 90 | 124.67 | 130.97 | 102.40 |
| 1.00 | 00 | 112.47 | 130.97 | 378.17 |
| 2.00 | 60 | 120.26 | 130.97 | 347.61 |
| 3.00 | 00 | 112.47 | 130.97 | 297.89 |
| 4.00 | 60 | 124.67 | 130.97 | 255.10 |
| 5.00 | 00 | 112.47 | 130.97 | 102.40 |

| DISTANCE FROM CENTER | TIME | TEMPERATURE | TEMPERATURE OF CONTAINER | TIME DELAY |
|----------------------|------|-------------|--------------------------|------------|
| 0 | 540 | 113 | 128 | 388 |
| 1 | 540 | 114 | 128 | 377 |
| 2 | 540 | 115 | 128 | 347 |
| 3 | 540 | 118 | 128 | 297 |
| 4 | 540 | 121 | 128 | 232 |
| 5 | 540 | 125 | 128 | 152 |
| 0 | 570 | 115 | 124 | 188 |
| 1 | 570 | 117 | 124 | 377 |
| 2 | 570 | 119 | 124 | 297 |
| 3 | 570 | 121 | 124 | 232 |
| 4 | 570 | 123 | 124 | 152 |
| 5 | 570 | 124 | 124 | 102 |
| 0 | 600 | 117 | 121 | 188 |
| 1 | 600 | 118 | 121 | 377 |
| 2 | 600 | 119 | 121 | 297 |
| 3 | 600 | 121 | 121 | 232 |
| 4 | 600 | 123 | 121 | 152 |
| 5 | 600 | 125 | 121 | 102 |
| 0 | 630 | 118 | 117 | 188 |
| 1 | 630 | 119 | 117 | 377 |
| 2 | 630 | 120 | 117 | 297 |
| 3 | 630 | 121 | 117 | 232 |
| 4 | 630 | 121 | 117 | 152 |
| 5 | 630 | 121 | 117 | 102 |
| 0 | 660 | 118 | 112 | 188 |
| 1 | 660 | 119 | 112 | 377 |
| 2 | 660 | 120 | 112 | 297 |
| 3 | 660 | 121 | 112 | 232 |
| 4 | 660 | 121 | 112 | 152 |
| 5 | 660 | 121 | 112 | 102 |
| 0 | 690 | 119 | 108 | 188 |
| 1 | 690 | 119 | 108 | 377 |
| 2 | 690 | 119 | 108 | 297 |
| 3 | 690 | 120 | 108 | 232 |
| 4 | 690 | 120 | 108 | 152 |
| 5 | 690 | 116 | 108 | 102 |

| DISTANCE FROM CENTER | TIME | TEMPERATURE | TEMPERATURE OF CONTAINER | TIME DELAY |
|----------------------|--------|-------------|--------------------------|------------|
| 0.00 | 720.00 | 119.72 | 104.30 | 388.40 |
| 1.00 | 720.00 | 119.80 | 104.00 | 378.17 |
| 2.00 | 720.00 | 119.87 | 104.00 | 347.89 |
| 3.00 | 720.00 | 119.90 | 104.00 | 297.81 |
| 4.00 | 720.00 | 119.98 | 104.00 | 232.19 |
| 5.00 | 720.00 | 119.98 | 104.00 | 152.40 |
| 0.00 | 750.00 | 119.83 | 99.56 | 378.17 |
| 1.00 | 750.00 | 119.87 | 99.56 | 347.89 |
| 2.00 | 750.00 | 119.90 | 99.56 | 297.81 |
| 3.00 | 750.00 | 119.98 | 99.56 | 232.19 |
| 4.00 | 750.00 | 119.98 | 99.56 | 152.40 |
| 5.00 | 750.00 | 119.98 | 99.56 | 102.40 |
| 0.00 | 780.00 | 119.59 | 99.20 | 388.40 |
| 1.00 | 780.00 | 119.59 | 99.20 | 378.17 |
| 2.00 | 780.00 | 119.59 | 99.20 | 347.89 |
| 3.00 | 780.00 | 119.59 | 99.20 | 297.81 |
| 4.00 | 780.00 | 119.59 | 99.20 | 232.19 |
| 5.00 | 780.00 | 119.59 | 99.20 | 152.40 |
| 0.00 | 810.00 | 119.28 | 99.99 | 388.40 |
| 1.00 | 810.00 | 119.28 | 99.99 | 378.17 |
| 2.00 | 810.00 | 119.28 | 99.99 | 347.89 |
| 3.00 | 810.00 | 119.28 | 99.99 | 297.81 |
| 4.00 | 810.00 | 119.28 | 99.99 | 232.19 |
| 5.00 | 810.00 | 119.28 | 99.99 | 152.40 |
| 0.00 | 840.00 | 119.16 | 99.99 | 388.40 |
| 1.00 | 840.00 | 119.16 | 99.99 | 378.17 |
| 2.00 | 840.00 | 119.16 | 99.99 | 347.89 |
| 3.00 | 840.00 | 119.16 | 99.99 | 297.81 |
| 4.00 | 840.00 | 119.16 | 99.99 | 232.19 |
| 5.00 | 840.00 | 119.16 | 99.99 | 152.40 |
| 0.00 | 870.00 | 119.05 | 99.00 | 388.40 |
| 1.00 | 870.00 | 119.05 | 99.00 | 378.17 |
| 2.00 | 870.00 | 119.05 | 99.00 | 347.89 |
| 3.00 | 870.00 | 119.05 | 99.00 | 297.81 |
| 4.00 | 870.00 | 119.05 | 99.00 | 232.19 |
| 5.00 | 870.00 | 119.05 | 99.00 | 152.40 |
| 0.00 | 900.00 | 118.88 | 87.00 | 388.40 |
| 1.00 | 900.00 | 118.88 | 87.00 | 378.17 |
| 2.00 | 900.00 | 118.88 | 87.00 | 347.89 |
| 3.00 | 900.00 | 118.88 | 87.00 | 297.81 |
| 4.00 | 900.00 | 118.88 | 87.00 | 232.19 |
| 5.00 | 900.00 | 118.88 | 87.00 | 152.40 |
| 0.00 | 930.00 | 118.75 | 87.00 | 388.40 |
| 1.00 | 930.00 | 118.75 | 87.00 | 378.17 |
| 2.00 | 930.00 | 118.75 | 87.00 | 347.89 |
| 3.00 | 930.00 | 118.75 | 87.00 | 297.81 |
| 4.00 | 930.00 | 118.75 | 87.00 | 232.19 |
| 5.00 | 930.00 | 118.75 | 87.00 | 152.40 |
| 0.00 | 960.00 | 118.63 | 87.00 | 388.40 |
| 1.00 | 960.00 | 118.63 | 87.00 | 378.17 |
| 2.00 | 960.00 | 118.63 | 87.00 | 347.89 |
| 3.00 | 960.00 | 118.63 | 87.00 | 297.81 |
| 4.00 | 960.00 | 118.63 | 87.00 | 232.19 |
| 5.00 | 960.00 | 118.63 | 87.00 | 152.40 |
| 0.00 | 990.00 | 118.51 | 87.00 | 388.40 |
| 1.00 | 990.00 | 118.51 | 87.00 | 378.17 |
| 2.00 | 990.00 | 118.51 | 87.00 | 347.89 |
| 3.00 | 990.00 | 118.51 | 87.00 | 297.81 |
| 4.00 | 990.00 | 118.51 | 87.00 | 232.19 |
| 5.00 | 990.00 | 118.51 | 87.00 | 152.40 |

| DISTANCE FROM CENTER | TIME | TEMPERATURE | TEMPERATURE OF CONTAINER | TIME DELAY |
|----------------------|---------|-------------|--------------------------|------------|
| 0.00 | 900.00 | 116.50 | 79.96 | 388.40 |
| 1.00 | 900.00 | 114.66 | 79.96 | 378.17 |
| 2.00 | 900.00 | 112.10 | 79.96 | 347.89 |
| 3.00 | 900.00 | 108.06 | 79.96 | 232.81 |
| 4.00 | 900.00 | 102.15 | 79.96 | 159.10 |
| 5.00 | 900.00 | 96.22 | 79.96 | 102.40 |
| 0.00 | 930.00 | 114.62 | 77.03 | 378.17 |
| 1.00 | 930.00 | 113.02 | 77.03 | 347.89 |
| 2.00 | 930.00 | 110.16 | 77.03 | 297.81 |
| 3.00 | 930.00 | 105.57 | 77.03 | 232.81 |
| 4.00 | 930.00 | 99.51 | 77.03 | 159.10 |
| 5.00 | 930.00 | 93.51 | 77.03 | 102.40 |
| 0.00 | 960.00 | 112.59 | 74.56 | 388.40 |
| 1.00 | 960.00 | 111.22 | 74.56 | 378.17 |
| 2.00 | 960.00 | 108.44 | 74.56 | 347.89 |
| 3.00 | 960.00 | 103.59 | 74.56 | 297.81 |
| 4.00 | 960.00 | 96.90 | 74.56 | 232.81 |
| 5.00 | 960.00 | 90.83 | 74.56 | 159.10 |
| 0.00 | 990.00 | 111.30 | 72.59 | 388.40 |
| 1.00 | 990.00 | 109.99 | 72.59 | 378.17 |
| 2.00 | 990.00 | 106.44 | 72.59 | 347.89 |
| 3.00 | 990.00 | 101.13 | 72.59 | 297.81 |
| 4.00 | 990.00 | 94.52 | 72.59 | 232.81 |
| 5.00 | 990.00 | 88.93 | 72.59 | 159.10 |
| 0.00 | 1020.00 | 109.30 | 71.16 | 388.40 |
| 1.00 | 1020.00 | 107.28 | 71.16 | 378.17 |
| 2.00 | 1020.00 | 103.81 | 71.16 | 347.89 |
| 3.00 | 1020.00 | 98.30 | 71.16 | 297.81 |
| 4.00 | 1020.00 | 92.49 | 71.16 | 232.81 |
| 5.00 | 1020.00 | 87.31 | 71.16 | 159.10 |
| 0.00 | 1050.00 | 107.30 | 70.29 | 388.40 |
| 1.00 | 1050.00 | 105.27 | 70.29 | 378.17 |
| 2.00 | 1050.00 | 101.69 | 70.29 | 347.89 |
| 3.00 | 1050.00 | 96.25 | 70.29 | 297.81 |
| 4.00 | 1050.00 | 90.84 | 70.29 | 232.81 |
| 5.00 | 1050.00 | 85.48 | 70.29 | 159.10 |

| DISTANCE FROM CENTER | TIME | TEMPERATURE | TEMPERATURE OF CONTAINER | TIME DELAY |
|----------------------|--------|-------------|--------------------------|------------|
| 0.00 | 080.00 | 96 | 70.00 | 38.47 |
| 1.00 | 080.00 | 105.26 | 70.00 | 37.16 |
| 2.00 | 080.00 | 103.54 | 70.00 | 47.81 |
| 3.00 | 080.00 | 94.64 | 70.00 | 22.81 |
| 4.00 | 080.00 | 88.41 | 70.00 | 51.09 |
| 5.00 | 080.00 | 83.18 | 70.29 | 22.47 |
| 0.00 | 110.00 | 103.18 | 70.29 | 37.16 |
| 1.00 | 110.00 | 107.96 | 70.29 | 47.81 |
| 2.00 | 110.00 | 101.72 | 70.29 | 22.81 |
| 3.00 | 110.00 | 92.55 | 70.29 | 51.09 |
| 4.00 | 110.00 | 81.87 | 70.29 | 22.47 |
| 0.00 | 140.00 | 101.82 | 71.16 | 38.47 |
| 1.00 | 140.00 | 109.10 | 71.16 | 47.81 |
| 2.00 | 140.00 | 95.63 | 71.16 | 22.81 |
| 3.00 | 140.00 | 81.27 | 71.16 | 51.09 |
| 4.00 | 140.00 | 75.29 | 71.16 | 22.47 |
| 0.00 | 170.00 | 99.11 | 72.59 | 38.47 |
| 1.00 | 170.00 | 97.89 | 72.59 | 47.81 |
| 2.00 | 170.00 | 89.57 | 72.59 | 22.81 |
| 3.00 | 170.00 | 84.53 | 72.59 | 51.09 |
| 4.00 | 170.00 | 80.84 | 72.59 | 22.47 |
| 0.00 | 200.00 | 97.12 | 74.56 | 38.47 |
| 1.00 | 200.00 | 95.28 | 74.56 | 47.81 |
| 2.00 | 200.00 | 84.34 | 74.56 | 22.81 |
| 3.00 | 200.00 | 80.09 | 74.56 | 51.09 |
| 4.00 | 200.00 | 77.31 | 74.56 | 22.47 |
| 0.00 | 230.00 | 95.36 | 77.03 | 38.47 |
| 1.00 | 230.00 | 93.08 | 77.03 | 47.81 |
| 2.00 | 230.00 | 85.31 | 77.03 | 22.81 |
| 3.00 | 230.00 | 81.73 | 77.03 | 51.09 |
| 4.00 | 230.00 | 78.11 | 77.03 | 22.47 |
| 0.00 | 260.00 | 93.11 | 77.03 | 38.47 |
| 1.00 | 260.00 | 91.83 | 77.03 | 47.81 |
| 2.00 | 260.00 | 83.11 | 77.03 | 22.81 |
| 3.00 | 260.00 | 79.73 | 77.03 | 51.09 |
| 4.00 | 260.00 | 76.31 | 77.03 | 22.47 |

| DISTANCE FROM CENTER | TIME | TEMPERATURE | TEMPERATURE OF CONTAINER | TIME DELAY |
|----------------------|---------|-------------|--------------------------|------------|
| 0.00 | 1260.00 | 94.27 | 79.96 | 388.47 |
| 1.00 | 1260.00 | 93.72 | 79.96 | 377.61 |
| 2.00 | 1260.00 | 92.66 | 79.96 | 397.89 |
| 3.00 | 1260.00 | 88.70 | 79.96 | 329.10 |
| 4.00 | 1260.00 | 83.15 | 79.96 | 102.29 |
| 5.00 | 1260.00 | 82.73 | 83.30 | 378.17 |
| 0.00 | 1290.00 | 92.53 | 83.30 | 378.17 |
| 1.00 | 1290.00 | 90.73 | 83.30 | 397.68 |
| 2.00 | 1290.00 | 86.22 | 83.30 | 297.81 |
| 3.00 | 1290.00 | 84.34 | 83.30 | 159.29 |
| 4.00 | 1290.00 | 83.36 | 83.30 | 102.40 |
| 5.00 | 1320.00 | 91.94 | 87.00 | 378.17 |
| 0.00 | 1320.00 | 89.76 | 87.00 | 347.69 |
| 1.00 | 1320.00 | 88.04 | 87.00 | 297.81 |
| 2.00 | 1320.00 | 86.29 | 87.00 | 229.10 |
| 3.00 | 1320.00 | 84.47 | 87.00 | 102.29 |
| 4.00 | 1350.00 | 89.88 | 90.99 | 378.17 |
| 5.00 | 1350.00 | 88.28 | 90.99 | 347.69 |
| 0.00 | 1350.00 | 86.65 | 90.99 | 297.81 |
| 1.00 | 1350.00 | 84.57 | 90.99 | 229.10 |
| 2.00 | 1350.00 | 82.05 | 90.99 | 102.29 |
| 3.00 | 1380.00 | 89.32 | 95.20 | 378.17 |
| 4.00 | 1380.00 | 87.32 | 95.20 | 347.69 |
| 5.00 | 1380.00 | 85.99 | 95.20 | 297.81 |
| 0.00 | 1380.00 | 83.59 | 95.20 | 229.10 |
| 1.00 | 1380.00 | 81.45 | 95.20 | 102.29 |
| 2.00 | 1410.00 | 88.37 | 99.56 | 378.17 |
| 3.00 | 1410.00 | 86.68 | 99.56 | 347.69 |
| 4.00 | 1410.00 | 84.77 | 99.56 | 297.81 |
| 5.00 | 1410.00 | 82.91 | 99.56 | 229.10 |
| 0.00 | 1440.00 | 89.13 | 99.56 | 378.17 |
| 1.00 | 1440.00 | 87.10 | 99.56 | 347.69 |
| 2.00 | 1440.00 | 84.91 | 99.56 | 297.81 |
| 3.00 | 1440.00 | 82.56 | 99.56 | 229.10 |
| 4.00 | 1440.00 | 81.10 | 99.56 | 102.29 |
| 5.00 | 1440.00 | 79.13 | 99.56 | 102.29 |

APPENDIX C

TRUMP Solution

TRUMP is a computer program for solving transient and steady-state temperature distributions in multidimensional systems. This program was developed in 1965 at the Lawrence Radiation Laboratory by A. L. Edwards [Ref. 2] for their CDC/3600 computer. The program was adapted to the Naval Postgraduate School IBM/360 Model 67 computer system in 1971 by C. Erbayrum [Ref. 3] from a version used by the B. F. Goodrich Corporation.

TRUMP is a multi-purpose program able to solve a wide variety of problems involving flow in various kinds of potential fields such as heat flow in a temperature field. TRUMP allows the solution of general nonlinear parabolic partial differential equations both in steady-state and transient problems. Complex geometric configurations with multidimensional flow may be solved using various coordinate systems. Initial conditions may vary with spatial position. Material properties, boundary conditions, and other problem parameters may vary with spatial position, time, or the primary dependent variable.

Input data are fed to TRUMP in "Block" form through its 12 input data blocks. A complete description of each of these blocks is given in Ref. 2. A model of the problem must be constructed and data from this model read into TRUMP through the data blocks.

Two models were used to simulate the rocket motor storage container system and several variations of each model were investigated.

The first model assumed one dimensional heat transfer (radial) with the assumptions that the system was infinitely long and that the container surface temperature was spatially uniform. The system was modeled as two infinitely long concentric cylinders separated by a 2.94 inch air gap. The inner cylinder was constructed of 4130 steel and was filled with dry wind blown sand. The thermal properties of the materials used in the experimental system are given in Table II with units most easily compared to the actual data obtained from the system at China Lake.

TABLE II
Thermal Properties of Materials

| Material | Density | Specific Heat | Thermal Conductivity |
|----------|-----------------------------|-----------------|-----------------------|
| Sand | 0.05486 lbm/in ³ | 0.195 BTU/lbm°F | 0.00026 BTU/min-in-°F |
| Steel | 0.2807 lbm/in ³ | 0.109 BTU/lbm°F | 0.364 BTU/min-in-°F |
| Air | 0.0000436 " | 0.240 BTU/lbm°F | 0.0000225 " |

The model was subdivided into volume elements or nodes with the representative nodal points given in Figure 27. Although the representative nodal point may be located anywhere in the node or on the surface of the node, in transient problems it is usually located so that the lines connecting the nodal points are perpendicularly bisected by the connected area. This gives maximum accuracy. Two boundary conditions were given to the surface node. The first was a sinusoidal disturbance which closely modeled the actual

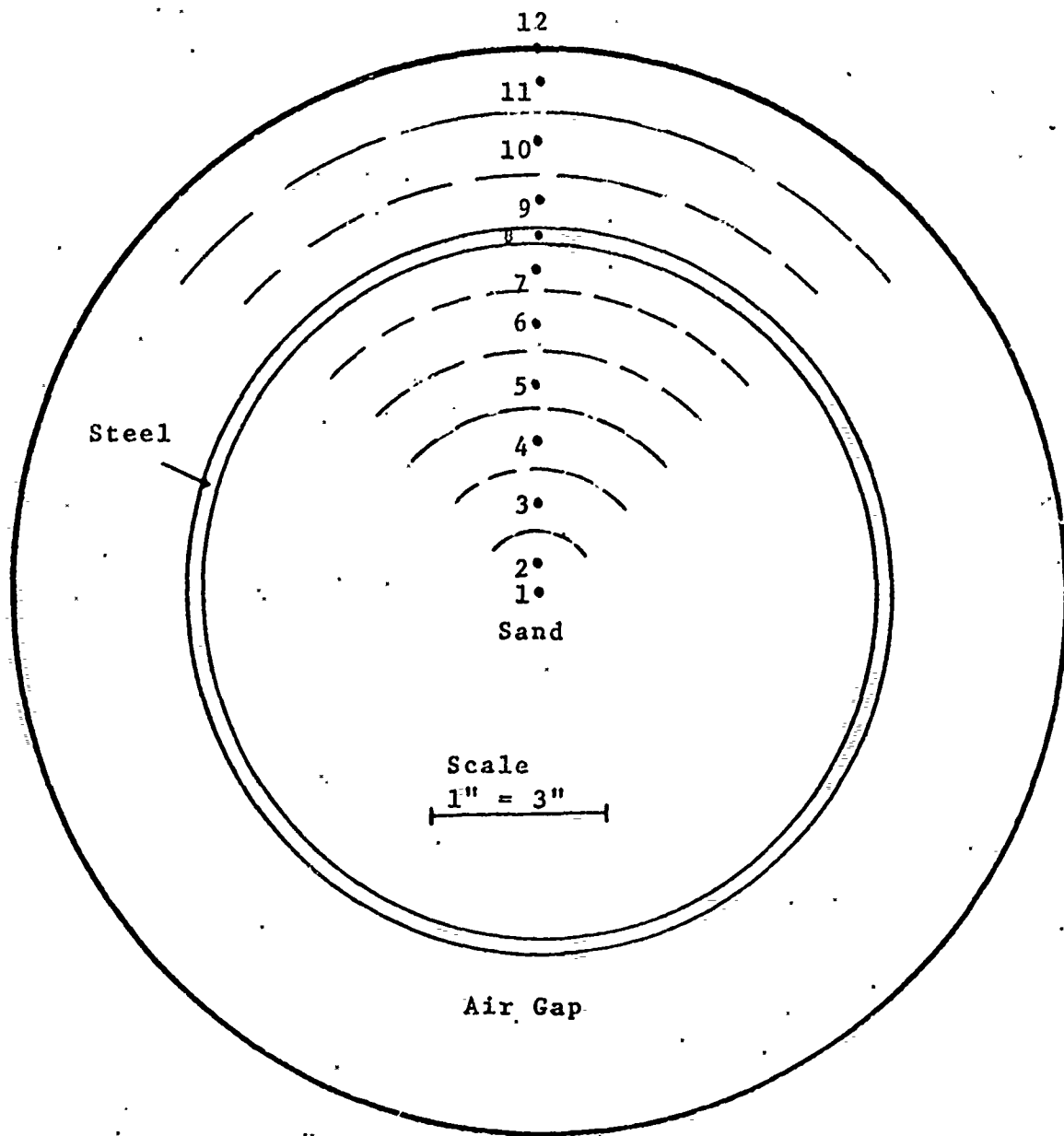


Figure 27. Location of Nodes for One Dimensional TRUMP Model.

average experimental data obtained at China Lake as given in Appendix D. The sine wave exhibited a maximum temperature of 138°F and an average temperature of 104°F. Its period was 24 hours (1440 minutes). The second boundary condition was the actual average surface temperature of the storage container given at two hour intervals. Both these boundary conditions are approximations of the actual surface temperature. Two hour intervals were the minimum allowable for the tabular data as this version of TRUMP has a maximum table size of 12.

Several assumptions were initially made. The thermocouple data obtained from the experiment at China Lake gave the average temperature at a point on the storage container and not the actual outside surface temperature. As this container wall was only 1/16 of an inch thick and made of a good thermal conductor, it was decided to model this data as a zero volume boundary node with a known temperature impressed on it. It was also assumed that heat transfer across the air gap occurred only by radiation and conduction, neglecting the effects of free convection.

It was estimated that the surface emissivities for the rocket motor and the storage container were 0.9 [Ref. 6] based on their haze gray surfaces. The radiation exchange factor for this geometry is given by

$$F_{1-2} = \frac{1}{\frac{1}{\epsilon_1} + \frac{r_1}{r_2} \left(\frac{1}{\epsilon_2} - 1 \right)} = 0.84$$

A sample input deck for the tabular approximation of the boundary condition is given at the end of this appendix. Several cycles of output data for the one dimensional model are also given.

The second model assumed two dimensional heat transfer (radial and circumferential) with the assumption that the system was infinitely long. The same physical model was assumed for the system except 48 nodes were used instead of 12. The representative nodal points are given in Figure 28. The four surface nodes (12, 24, 36, and 48) each had two different temperature approximations applied, a sinusoidal representation and a tabular input taken at two hour intervals. The four surface nodes were also modeled as zero volume boundary nodes. Each internal thermal connection between nodes is described in the input data by specifying the two node identification numbers, two connector lengths, and two interface dimensional factors. An example of the thermal connections of node 4 is shown in Figure 28 and the input data in BLOCK 5 of the two dimensional TRUMP program.

The calculation of the radiosities in the two dimensional case was accomplished by using a radiation-network and the method of crossed-strings.

The radiation shape factors for the two dimensional system were determined by the method of crossed-strings [Ref. 14]. The graphical construction for this method is given in Figure 29.

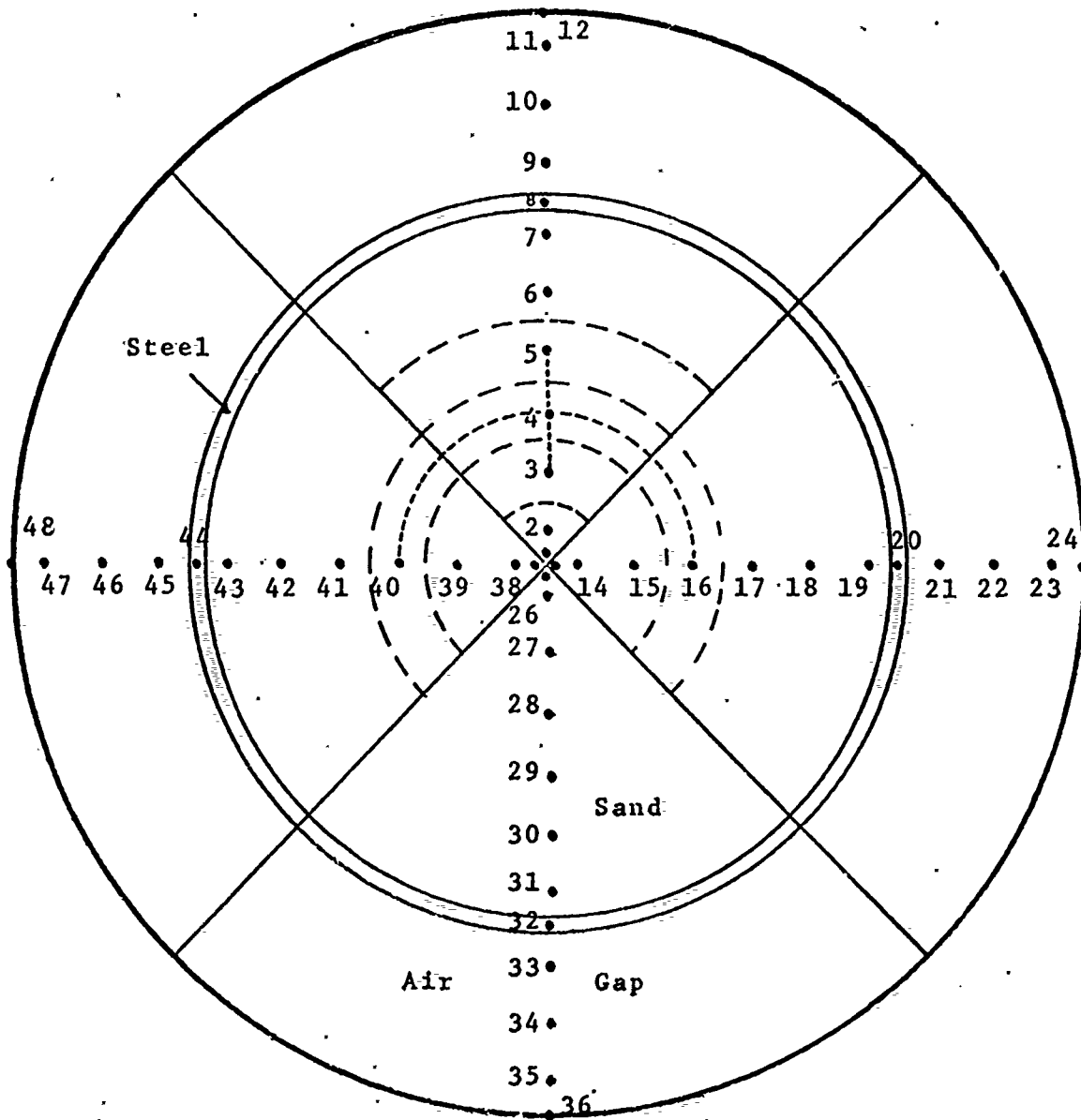


Figure 28: Location of Nodes for Two Dimensional TRUMP Model.

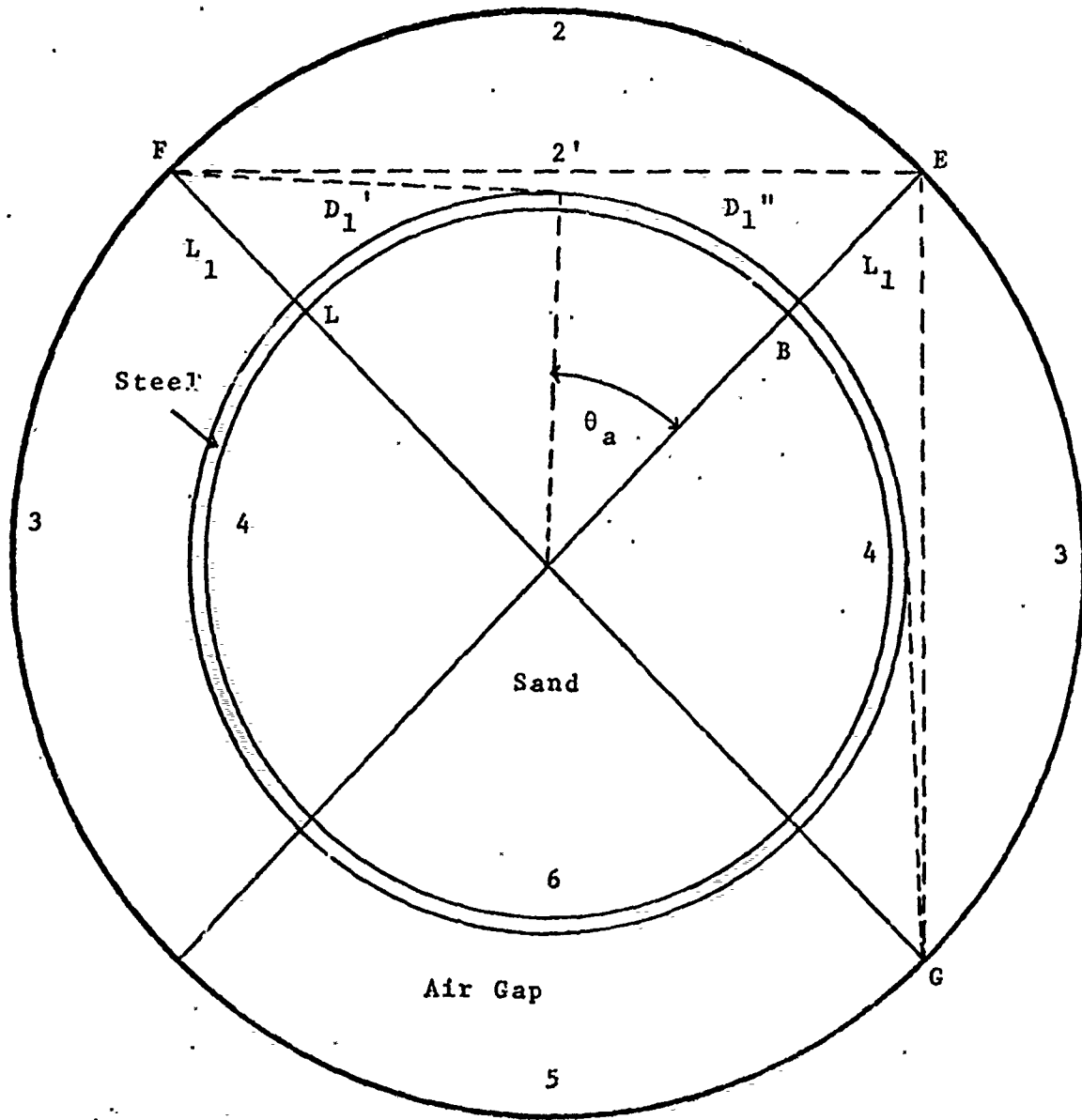


Figure 29: Graphical Construction for Crossed-Strings Method.

F_{m-n} is defined as the fraction of energy leaving surface m which directly reaches surface n . From the physical dimensions of the model $A_1 = A_4 = A_6 = 9.45 \frac{\text{sq.in}}{\text{in.}}$, $A_2 = A_3 = A_5 = 14.05 \frac{\text{sq.in}}{\text{in.}}$, $A_{2'} = 12.65 \frac{\text{sq.in}}{\text{in.}}$, assuming unit depth. Let S_i equal the length of A_i .

The crossed-string method lets each surface represent the effective surface obtained by stretching a string tightly over the radiating face between the bounding edges, to produce a surface that cannot see any of itself. For example, surface $2'$ in Figure 29 stretched over surface 2. By definition $F_{2',2} \equiv 1$, which by reciprocity leads to

$$F_{2,2'} = \frac{A_{2'}}{A_2} F_{2',2} = 0.9$$

and therefore since

$$F_{2,2} + F_{2,2'} = 1 \quad \text{then } F_{2,2} = 0.10$$

To calculate the direct radiant heat exchanged between surfaces 1 and 2, a minimum-length line was stretched connecting edge B of A_1 to edge E of A_2 and a second minimum length line from edge L of A_1 to edge F of A_2 . These lines are labeled L_1 in Figure 29 and are equal to the width of the air gap, $L_1 = 2.9375$ in. Minimum length lines were also stretched from point B on A_1 to F on A_2 and L on A_1 to E on A_2 . The length of these lines is D_1 and is made up of two parts; D_1' , the tangential distance from F to surface A_1 and D_1'' , the arc length from the point the tangent hits A_1 to B. From geometry $D_1' = \sqrt{r_1^2 - r_0^2} = 6.62''$

$$D_1'' = r_0 \theta_a = 4.42''$$

therefore

$$D_1 = D_1' + D_1'' = 11.04''$$

$$\text{Now } F_{12} = \frac{2D_1 - 2L_1}{2S_1} = 0.86$$

From reciprocity, $A_1 F_{12} = A_2 F_{21}$

$$F_{21} = \frac{r_0}{r_1} F_{12} = 0.578$$

Now F_{13} is calculated from

$$F_{12} + 2F_{13} = 1$$

therefore $F_{13} = 0.07$

From symmetry $F_{42} = F_{13}$

and then by reciprocity

$$F_{24} = \frac{r_0}{r_1} F_{42} = 0.047$$

Now F_{23} was calculated by stretching minimum length lines from F to E, from E to G, from F to G and from E to E; where the length of the line from F to E = from E to G = S_2' , from F to G = $2D_1$, and from E to E = 0

$$F_{23} = \frac{2S_2' - 2D_1}{2S_2} = .113$$

F_{25} was now calculated from

$$F_{21} + F_{22} + F_{23} + F_{24} + F_{25} = 1$$

therefore $F_{25} = 0.002$

As F_{25} was much smaller than the other F_{m-n} , it was not included in the radiation-network diagram in Figure 30

[Ref. 15].

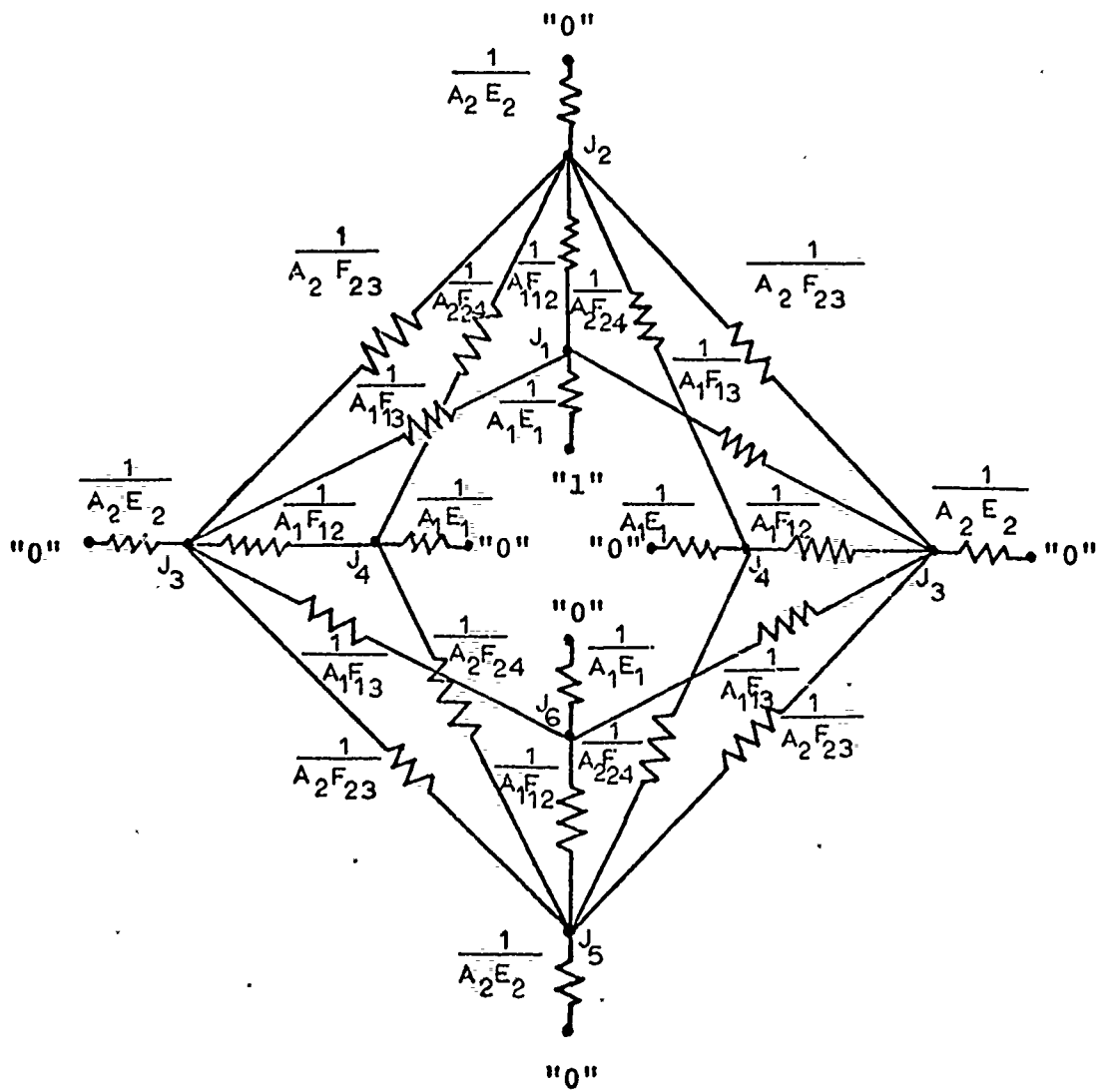


Figure 30: Radiation Network.

To calculate F_{1-n} , the blackbody potential of area 1 is set to unity and all other blackbody potentials are set as zero. An energy balance was written at each node giving a set of six simultaneous equations as follows:

Node 1

$$E_1 A_1 (1 - J_1) = A_1 F_{12} (J_1 - J_2) + 2 A_1 F_{13} (J_1 - J_3)$$

Node 2

$$E_2 A_2 (0 - J_2) = A_1 F_{12} (J_2 - J_1) + 2 A_2 F_{24} (J_2 - J_4) + 2 A_2 F_{23} (J_2 - J_3)$$

Node 3

$$E_2 A_2 (0 - J_3) = A_1 F_{12} (J_3 - J_4) + A_2 F_{23} (J_3 - J_2) + A_1 F_{13} (J_3 - J_1) + A_1 F_{13} (J_3 - J_6) + A_2 F_{23} (J_3 - J_5)$$

Node 4

$$E_1 A_1 (0 - J_4) = A_2 F_{24} (J_4 - J_2) + A_1 F_{12} (J_4 - J_3) + A_2 F_{24} (J_4 - J_5)$$

Node 5

$$E_2 A_2 (0 - J_5) = 2 A_2 F_{23} (J_5 - J_3) + 2 A_2 F_{24} (J_5 - J_4) + A_1 F_{12} (J_5 - J_6)$$

Node 6

$$E_1 A_1 (0 - J_6) = 2 A_1 F_{13} (J_6 - J_3) + A_1 F_{12} (J_6 - J_5)$$

where J_n = radiosity of node n .

$$E_1 A_1 = \frac{\epsilon_1}{1 - \epsilon_1} A_1$$

$$E_2 A_2 = \frac{\epsilon_2}{1 - \epsilon_2} A_2$$

F_{nm} = radiation shape factors previously calculated.

Now the values of A_1 , A_2 and F_{nm} were substituted into the energy balance equations which were then put into matrix form as shown in Table III.

TABLE III
Matrix Form of Energy Balance Equations

| | | | | | | |
|---------------------------------------|---|---|---------------------------------------|---|---------------------------------------|--|
| $\left(\frac{1}{1-\epsilon_1}\right)$ | (-0.86) | (-0.14) | (0.0) | (0.0) | (0.0) | $\left(\frac{\epsilon_1}{1-\epsilon_1}\right)$ |
| (-0.86) | $\left(1.33 + \frac{3\epsilon_2}{2(1-\epsilon_2)}\right)$ | (-0.33) | (-0.14) | (0.0) | (0.0) | (0.0) |
| (-0.07) | (-0.165) | $\left(1.33 + \frac{3\epsilon_2}{2(1-\epsilon_2)}\right)$ | (-0.86) | (-0.165) | (-0.07) | (0.0) |
| (0.0) | (-0.07) | (-0.86) | $\left(\frac{1}{1-\epsilon_1}\right)$ | (-0.07) | (0.0) | (0.0) |
| (0.0) | (0.0) | (-0.14) | (0.0) | (-0.86) | $\left(\frac{1}{1-\epsilon_1}\right)$ | (0.0) |
| (0.0) | (0.0) | (-0.33) | (-0.14) | $\left(1.33 + \frac{3\epsilon_2}{2(1-\epsilon_2)}\right)$ | (-0.86) | (0.0) |

Letting $\epsilon_1 = \epsilon_2 = 0.9$, a standard computer solution for matrix problems gave the radiosities as listed in Table IV.

TABLE IV
Radiosities at Nodes

| | | |
|-------|---|---------|
| J_1 | = | .9046 |
| J_2 | = | .0529 |
| J_3 | = | .00495 |
| J_4 | = | .000797 |
| J_5 | = | .000125 |
| J_6 | = | .000080 |

Now to find the radiation exchange factors from node 1 to nodes n, the radiation network shown in Figure 30 was reduced to the equivalent network shown in Figure 31. Where the nodal equations are

$$F_{1-2} A_1 (1-\epsilon) = \epsilon_2 A_2 (J_2 - J_1)$$

where

$$F_{1-2} = \frac{\epsilon_2 A_2}{A_1} J_2 = .71$$

$$F_{1-3} A_1 (1-\epsilon) = \epsilon_2 A_2 (J_3 - J_1)$$

$$F_{1-3} = \frac{\epsilon_2 A_2}{A_1} J_3 = .066$$

These values of the radiation exchange factor are used in the two-dimensional program. A sample input deck for the sinusoidal boundary condition is included at the end of this appendix. Several cycles of output data for the tabular boundary condition are also given.

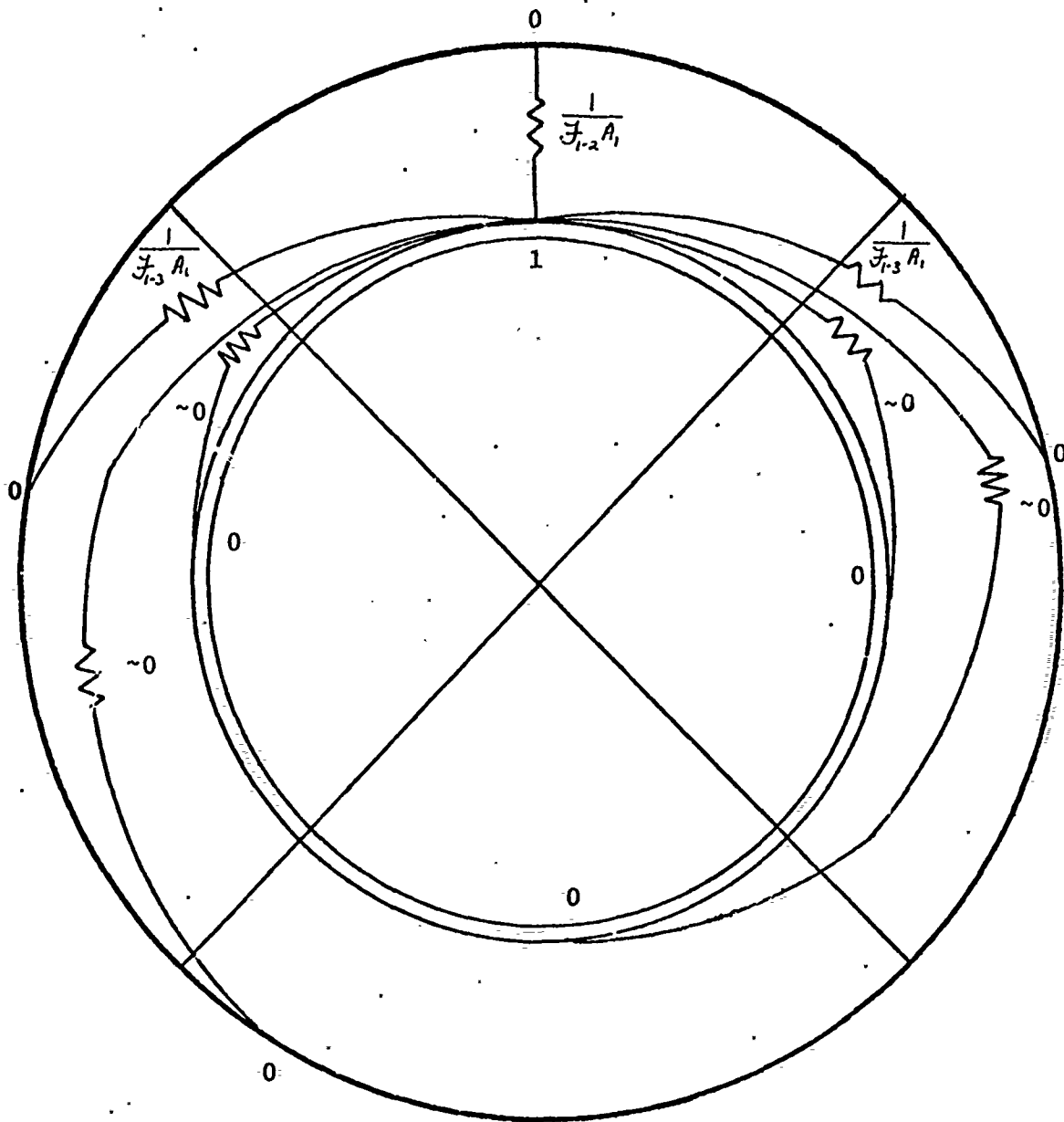


Figure 31: Equivalent Radiation Network

```

//WIR31687 JOB (1687,0860FT,NF12),WIRZBURGER,A. BOX W',TIME=(2,00)
//JOB18 DD UNIT=2321,DSNAME=SI734.KATZ,
// DISP=(OLD,PASS),VOLUME=SER=CELOO1
// EXEC PGM=TRUMP,REGION=350K
//FT06FOO1 DD SYSOUT=A,DCB=(RECFM=FBA,LRECL=133,BLKSIZE=3325),
//FT05FOO1 DD SPACE=(CYL,(5,1))

```

* MISSILE PROBLEM ONE DIMENSIONAL

```

BLOCK 1 CONTROLS, LIMITS, CONSTANTS
1 1.000 E 00 1.000E 00 1440.0
93.0

```

```

BLOCK 2 MATERIAL NAMES, NUMBERS, THERMAL PROPERTIES
ASAND 1 0.0544 0.19 0.00027
ASTEL 2 0.2807 0.109 0.0364
AAIR 3 0.0000436 0.240 0.0000225

```

```

BLOCK 4 NODE NUMBERS, MATERIAL REFERENCES, TYPES, VOLUMES.
1 1.0 0.002 0.001
2 1.0 0.998 0.501
3 1.0 1.0 1.5
4 1.0 1.0 3.5
5 1.0 1.0 4.5
6 1.0 0.75 5.375
7 1.0 0.25 5.875
8 1.0 1.0 6.5
9 1.0 0.9375 7.5
10 1.0 0.0 8.46875
11 1.0 0.0 8.9375
12 1.0 0.0 8.9375

```

```

BLOCK 5 INTERNAL THERMAL CONNECTION NODE NUMBERS.
1 0.001 0.499 0.002
2 0.499 0.5 1.0
3 0.5 0.5 1.0
4 0.5 0.5 1.0
5 0.5 0.375 1.0
6 0.375 0.125 1.0
7 0.125 0.5 1.0
8 0.5 0.5 6.0
9 0.5 0.5 6.0
10 0.5 0.5 7.0
11 0.5 0.5 8.0
12 0.46875 0.0 8.9375

```

0.00009760

| | | | | | | |
|---------|--------------------------------|--------|--------|--------|--------|--------|
| BLOCK 6 | EXTERNAL THERMAL CONNECTIONS | 1.0 | 8.9375 | 1.00 | E6 | |
| 12 2001 | | | | | | |
| BLOCK 7 | BOUNDARY TEMPERATURE VARIATION | 120.0 | 84.5 | 240.0 | 79.25 | 360.0 |
| 2001 | | 600.0 | 82.25 | 720.0 | 103.75 | 840.0 |
| 76.0 | | 1080.0 | 138.0 | 1200.0 | 132.25 | 1320.0 |
| 119.0 | | | | | | |
| 103.5 | | | | | | |
| | | | | | | |
| BLOCK 9 | INITIAL TEMPERATURES | | | | | |
| 1 | | 115.2 | | | | |
| 2 | | 115.3 | | | | |
| 3 | | 116.0 | | | | |
| 4 | | 117.2 | | | | |
| 5 | | 118.4 | | | | |
| 6 | | 118.7 | | | | |
| 7 | | 117.3 | | | | |
| 8 | | 115.8 | | | | |
| 9 | | 113.2 | | | | |
| 10 | | 108.9 | | | | |
| 11 | | 105.1 | | | | |
| 12 | | 103.5 | | | | |

ENDED-1 LAST CARD OF DATA DECK

MISSILE PROBLEM ONE DIMENSIONAL
 INPUT UNIT = 5. OUTPUT UNIT = 6.
 DATA BLOCK 10 CONTROLS, LIMITS, CONSTANTS

IPRINT NUM KDATA KSPEC MCYC MSEC NPUNCH NDOOT IRITE 0 SCALE
 15 0 1 0 30000 30000 0 0 0.10000E 01
 KD KT 3 DELTO 1.00000E 12 SMALL' TVARY 1.00000E 00 TAU 1.44000E 03 TMIN -1.00000E 12 TMAX 1.00000E 12
 KD KSYM GEOM SIGMA TRASE TRASE 4.60000E 02

TCNE 9.2000E 01 ALCNE 0.0 BONE 0.0 GONE 0.0 FONE 0.0 HGONE 0.0 RONE 0.0 PONE 0.0
 DATA BLOCK 20 MATERIAL NAMES, NUMBERS, THERMAL PROPERTIES.

| NAME | MATL | INDEX | KA | KB | LTAB | LTAB | DENSITY | CAPACITY | CONDUCTIVITY | TMELT | HMELT |
|------|------|-------|----|----|------|------|------------|-----------|--------------|-------|-------|
| SAND | 1 | 0 | 0 | 0 | 0 | 0 | 5.400E-02 | 1.900E-01 | 2.700E-04 | 0.0 | 0.0 |
| STEL | 2 | 0 | 0 | 0 | 0 | 0 | 2.8070E-01 | 1.090E-01 | 3.640E-02 | 0.0 | 0.0 |
| AIR | 3 | 0 | 0 | 0 | 0 | 0 | 4.360E-05 | 2.400E-01 | 2.250E-05 | 0.0 | 0.0 |

DATA BLOCK 40 NODE NUMBERS, MATERIAL REFERENCES, TYPES, VOLUMES.

| NODE | INDEX | MATL | NTYPE | DLONG | DWIDE | DRAD | HINT | RINT | VOLUME |
|------|-------|------|-------|-------------|-------------|-------------|-------------|-------------|-------------|
| 1 | 1 | 1 | 1 | 1.00000E 00 | 2.00000E-03 | 1.00000E-03 | 1.00000E-01 | 1.00000E 00 | 1.25664E-05 |
| 2 | 2 | 1 | 1 | 1.00000E 00 | 9.98000E-01 | 1.50000E-01 | 1.00000E 00 | 1.00000E 00 | 1.12197E 00 |
| 3 | 3 | 1 | 1 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.22478E 00 |
| 4 | 4 | 1 | 1 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 9.15201E 01 |
| 5 | 5 | 1 | 1 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.18971E 01 |
| 6 | 6 | 1 | 1 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 2.25273E 01 |
| 7 | 7 | 1 | 1 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 2.25273E 01 |
| 8 | 8 | 1 | 1 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 9.22543E 01 |
| 9 | 9 | 1 | 1 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 4.09407E 01 |
| 10 | 10 | 1 | 1 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 4.71239E 01 |
| 11 | 11 | 1 | 1 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E 00 | 4.98850E 01 |
| 12 | 12 | 1 | 1 | 1.00000E 00 | 9.37500E-01 | 8.93750E 00 | 1.00000E 00 | 1.00000E 00 | 1.00000E-24 |

DATA BLOCK 50 INTERNAL THERMAL CONNECTION NODE NUMBERS.

| NOD1 | NOD2 | INDEX | DEL1 | DEL2 | DLONG | DRAD | HINT | RINT | AREA |
|------|------|-------|------------|------------|------------|------------|------------|------------|------------|
| 1 | 2 | 1 | 1.000D-03 | 4.5900D-01 | 1.0000E 00 | 1.0000E-03 | 1.0000E 00 | 1.0000E 00 | 1.2566D-02 |
| 2 | 3 | 1 | 1.9900D-01 | 5.0000D-01 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 6.2032D 00 |
| 3 | 4 | 2 | 2.0000D-01 | 5.0000D-01 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 1.2566D 01 |
| 4 | 5 | 3 | 3.0000D-01 | 5.0000D-01 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 1.8850D 01 |
| 5 | 6 | 4 | 5.0000D-01 | 3.7500D-01 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 2.5133D 01 |
| 6 | 7 | 5 | 3.0000D-01 | 3.7500D-01 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 3.1416D 01 |
| 7 | 8 | 6 | 3.7500D-01 | 1.2500D-01 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 3.6128D 01 |
| 8 | 9 | 7 | 1.2500D-01 | 5.0000D-01 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 3.7699D 01 |
| 9 | 10 | 8 | 1.2500D-01 | 5.0000D-01 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 3.7699D 01 |
| 10 | 11 | 9 | 5.0000D-01 | 4.6875D-01 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 4.3982D 01 |
| 11 | 12 | 10 | 5.0000D-01 | 4.6875D-01 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 1.0000E 00 | 5.0245D 01 |
| 12 | 12 | 12 | 4.6875D-01 | 0.0 | 1.0000E 00 | 8.9375E 00 | 1.0000E 00 | 1.0000E 00 | 5.6156D 01 |

DATA BLOCK 60 EXTERNAL THERMAL CONNECTIONS

| NODS | NODSB | INDEX | LTAB | POWER | DLONG | DRAD | HSURE | RSURE | AREAS |
|------|-------|-------|------|-------|------------|------------|------------|-------|------------|
| 12 | 2001 | 1 | 0 | 0.0 | 1.0000E 00 | 8.9375E 00 | 1.0000D 06 | 0.0 | 5.6156D 01 |

DATA BLOCK 70

BOUNDARY TEMPERATURE VARIATION.

```

=====
NODB INDEX LTABT
2001      1      12
=====
      TEMPB      SLOPE      TIMER
      8.950000E 01 1.200000E 02
      8.450000E 01 4.370000E 02
      7.925000E 01 2.700000E 02
      7.600000E 01 4.800000E 02
      6.975000E 01 7.900000E 02
      6.225000E 01 1.200000E 02
      5.750000E 01 7.400000E 02
      5.150000E 02 1.770000E 01
      4.300000E 02 9.100000E 03
      3.580000E 02 6.160000E 02
      3.225000E 02 4.790000E 02
      1.035000E 02 2.395000E 01
=====
INITIAL TEMPERATURES
=====

```

NOTE INDEX

```

1 1.152000E 02 TT
2 1.1160000E 02 AA
3 1.1172000E 02 BB
4 1.1184000E 02 GG
5 1.1187000E 02
6 1.1173000E 02
7 1.1158000E 02
8 1.1132000E 02
9 1.0890000E 02
10 1.051000E 02
11 1.035000E 02
12
=====

```

DATA ENDED -10

LAST CARD OF DATA DECK

MACHINE TIME () = 0 SECONDS

SUMMARY OF INPUT DATA

```

BLOC NUMBER 2
TIME READ IN 1
ITEM NAME MAT 1
MAXIMUM SIZE 15
UNMODIFIED SIZE 0
CAPT CONT QT ZT ET HURTEMPB GT FLOWT TOTAL
0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0

```

TABLES

MAXIMUM ALLOWED TABLE LENGTH IS 12.

ARRAY STORAGE = 3*M11+M1*(1+M11)+M2*(11+7*M9)+M3*(5+9*M9)+56*M4+12*M5+M6*(12+3*M9)+M7*(5+3*M9)+M8*(3+3*M9)+M10*(9+3*M9)+5*M12+3*M9 = 39867.

OTHER TOTALS NOSPEC NOGEN VORAD NORADS NMELT NREACT
 1 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0

MATERIAL SUMMARY

```

NAME MATL NODES DENSITY CAPACITY TOT VOL TOT MASS TOT CAP TOT HEAT
SAND 1 7 5.44000E-02 1.90000E-01 1.03869E 02 5.65046E 00 1.07359E 00 1.26383E 02
STEL 2 2 2.80700E-01 1.09000E-01 9.22843E 00 2.59042E 00 2.82356E-01 3.26696E 01
AIR 3 3 4.36000E-05 2.40000E-01 1.37850E 02 6.01024E-03 1.44246E-03 1.56938E-01
SYSTEM TOTAL 7 12 2.50947E 02 8.24689E 00 1.35739D 00 1.59237E 02

```



```

TRUMP OUTPUT DATA
* MISSILE PROBLEM ONE DIMENSIONAL
PRINTOUT CYCLE TOO FAST TOO SLOW KMIT DELTIX 12 SMALL TVARY NUTS
4 30 0 1.00000E 00 1.00000E 00 1.00000E 00
TOTAL TIME 1.54166E 01 HEAT FLOW 7.70604E 02 TEMP RATE 2.38003E 00
2.38531E 02 5.67711E 02 3.23062E 00
AVG TEMP 1.04224E 02 HEAT CAPACITY 1.35739D 00 GEN RATE 0.0 HEAT GEN TEMP FROM GEN
CURE AT 280 F
NODE 01 1.126D 03 -0.7910D 00 -0.4074D -01 0.0 0.1483E -04 H 0.3379E -06 F 0.3379E -06
02 1.126D 03 -0.8245D 00 -0.4477D -01 0.0 0.3666E 01 0.3755E -01 0.3755E -01
03 1.107D 03 -0.8775D 00 -0.4919D -01 0.0 0.1089E 02 0.4708E 00 0.4708E 00
04 1.045D 03 -0.9267D 00 -0.5463D -01 0.0 0.2747E 01 0.5244E 01 0.5244E 01
05 1.045D 03 -0.9733D 00 -0.6007D -01 0.0 0.5074E 02 0.5744E 01 0.5744E 01
06 0.983D 03 -0.9733D 00 -0.6551D -01 0.0 0.2624E 02 0.4446E 01 0.4446E 01
07 0.983D 03 -0.9733D 00 -0.7155D 00 0.0 0.2776E 02 0.4544E 01 0.4544E 01
08 0.954D 02 -0.8935D 00 -0.4602D -01 0.0 0.4739E -01 0.4589E -02 0.4589E -02
09 0.954D 02 -0.8935D 00 -0.4602D -01 0.0 0.4465E -01 0.4465E -01 0.4465E -01
10 0.863D 02 -0.8231D 00 -0.4239D -01 0.0 0.2587E -02 0.2587E -02 0.2587E -02
11 0.863D 02 -0.8231D 00 -0.4239D -01 0.0 0.2587E -02 0.2587E -02 0.2587E -02
12 0.8456D 02 -0.8093D 00 -0.4167D -01 0.0 0.2587E -02 0.2587E -02 0.2587E -02
MATERIAL DATA
NAME MATL TOT CAP TOT HEAT TOT HEAT TOT HEAT HMELT
SAND 1 1.07359E 00 1.13585E 02 1.05800E 02 0.0 0.0
STFL 2 2.82356E -01 2.7569E 01 9.8393E 01 0.0 0.0
AIR 3 1.44246E -03 1.50534E -01 9.2443E 01 0.0 0.0
NODE DATA
NODE MATL NTYPE VOLUME MASS CAPACITY CONDUCTIVITY ZIP SLIM
1 1 4 0.1000E -02 0.6836D -06 0.1599D -06 0.2700D -03 0.6789E -05 0.1930D 01
2 1 4 0.1500E 00 0.4790D 00 0.3747E 00 0.5700D -03 0.5700E 00 0.1930D 01
3 1 4 0.1500E 01 0.8547D 00 0.7244D 00 0.2700D -03 0.8820E 01 0.1930D 01
4 1 4 0.2200E 01 0.1190D 01 0.2730D 00 0.2700D -03 0.1188E 01 0.1730D 02
5 1 4 0.2200E 01 0.1538D 01 0.2222D 00 0.2700D -03 0.1448E 01 0.1730D 02
6 1 4 0.5875E 01 0.2590D 01 0.2824D 00 0.2700D -03 0.3644E 01 0.1730D 02
7 1 4 0.5500E 01 0.9228E 02 0.4284E 02 0.2700D -04 0.3722E 01 0.1730D 02
8 1 4 0.5500E 01 0.7500E 01 0.4284E 02 0.2700D -04 0.3722E 01 0.1730D 02
9 1 4 0.5500E 01 0.5775D -02 0.4284E 02 0.2700D -04 0.3722E 01 0.1730D 02
10 1 4 0.8939E 01 0.2807D -04 0.3600D -05 0.2664D -01 0.3616D 08 0.1000D -23
INTERNAL CONNECTION DATA
NOD1 NOD2 AREA HINTO0E PINT HEAT FLOW AVG RATE
1 2 2 0.1235D 00 0.1000E 13 0.0 0.1628D -02 0.144E -08
2 3 3 0.1888D 02 0.1000E 13 0.0 0.3930D -02 0.269E -03
3 4 5 0.1888D 02 0.1000E 13 0.0 0.5089D -02 0.6556E -02
5 6 7 0.3144D 02 0.1000E 13 0.0 0.9540D -02 0.144E -01
6 7 8 0.3613D 02 0.1000E 13 0.0 0.9540D -02 0.144E -01
7 8 9 0.3770D 02 0.1000E 13 0.0 0.9540D -02 0.144E -01
8 9 10 0.4390D 02 0.1000E 13 0.0 0.9540D -02 0.144E -01
9 10 11 0.5020D 02 0.1000E 13 0.0 0.9540D -02 0.144E -01
10 11 12 0.5616D 02 0.1000E 13 0.0 0.9540D -02 0.144E -01
BOUNDARY NODE DATA
NOD0 TEMPR HEAT FLCH AVG RATE
2001 8.4561E 01 7.7060D 02 3.2306E 00
SYSTEM TOTAL 7.7060E 02 3.2306E 00
EXTERNAL CONNECTION DATA
NOD0 NOD05 AREAS HSURE HSURE POWER RSURE TRANS HEAT FLOW AVG RATE
2001 5.4561D 01 1.0000D 06 0.0 0.0 5.6156D 07 7.7060E 02 3.2306E 00

```

TRUMP OUTPUT DATA

```

***** MISSILE PROBLEM ONE DIMENSIONAL *****
PRINTOUT CYCLE 45 TCO FAST 6 TCO SLOW 0 DELTIX 1.00000E 12 SMALL 1.00000E 00 TVARY 1.00000E 00 MUTS 13
TOTAL TIME 5.51874E 02 TIME STEP 1.78817E 01 HEAT FLOW TCO 2.26636E 03 TEMP FROM FLUX 4.13667E 00 FLUX RATE 3.52543E 00
AVG TEMP 9.06828E 01 HEAT CAPACITY 1.35730E 00 HEAT CONTENT 1.23092E 02 GEN RATE 0.0 TEMP FROM GEN 3.0 TEMP RATE 3.0
CURE AT 280 F

```

```

***** MATERIAL DATA *****

```

| NAME | MATL | TOT CAP | TOT HEAT | AVG TEMP | TMELT | HMELT |
|-------|------|-------------|-------------|-------------|-------|-------|
| STEEL | 1 | 2.87266E-03 | 9.23118E 01 | 8.53166E 01 | 0.0 | 0.0 |
| AIR | 3 | 1.44246E-03 | 1.12364E-01 | 7.78977E 01 | 0.0 | 0.0 |

INTERNAL CONNECTION DATA

| NODE | MATL | NTYPE | RADIUS | AREA | PERIM | VOLUME | MASS | CAPACITY | CONDUCTIVITY | SLIM |
|------|------|-------|------------|------------|------------|------------|------------|------------|--------------|------------|
| 1 | 1 | 4 | 0.0000E-02 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 |
| 2 | 1 | 4 | 0.0000E-02 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 |
| 3 | 1 | 4 | 0.0000E-02 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 |
| 4 | 1 | 4 | 0.0000E-02 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 |
| 5 | 1 | 4 | 0.0000E-02 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 |
| 6 | 1 | 4 | 0.0000E-02 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 |
| 7 | 1 | 4 | 0.0000E-02 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 |
| 8 | 1 | 4 | 0.0000E-02 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 |
| 9 | 1 | 4 | 0.0000E-02 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 |
| 10 | 1 | 4 | 0.0000E-02 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 |
| 11 | 1 | 4 | 0.0000E-02 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 |
| 12 | 1 | 4 | 0.0000E-02 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 | 0.0000E-01 |

BOUNDARY NODE DATA

| NODE | TEMP | HEAT FLOW | AVG RATE | TRANS | RSURE | POWER | HSURE | AREAS | PERIM | MASS | CAP | CONDUCT | SLIM | AVG RATE |
|------|------------|-------------|------------|-------|-------|-------|-------|------------|------------|------------|-----|---------|------|----------|
| 2001 | 7.2257E 01 | 2.26640E 03 | 4.1067E 00 | 0.0 | 0.0 | 0.0 | 0.0 | 5.6156D 07 | 2.2664E 03 | 4.1067E 00 | 0.0 | 0.0 | 0.0 | 0.0 |

```

SYSTEM TOTAL
EXTERNAL CONNECTION DATA
NODS NODS01 AREAS 01 1.0000D 06 0.0
NODS NODS02 AREAS 02 1.0000D 06 0.0
NODS NODS03 AREAS 03 1.0000D 06 0.0
NODS NODS04 AREAS 04 1.0000D 06 0.0
NODS NODS05 AREAS 05 1.0000D 06 0.0
NODS NODS06 AREAS 06 1.0000D 06 0.0
NODS NODS07 AREAS 07 1.0000D 06 0.0
NODS NODS08 AREAS 08 1.0000D 06 0.0
NODS NODS09 AREAS 09 1.0000D 06 0.0
NODS NODS10 AREAS 10 1.0000D 06 0.0
NODS NODS11 AREAS 11 1.0000D 06 0.0
NODS NODS12 AREAS 12 1.0000D 06 0.0
NODS NODS13 AREAS 13 1.0000D 06 0.0
NODS NODS14 AREAS 14 1.0000D 06 0.0
NODS NODS15 AREAS 15 1.0000D 06 0.0
NODS NODS16 AREAS 16 1.0000D 06 0.0
NODS NODS17 AREAS 17 1.0000D 06 0.0
NODS NODS18 AREAS 18 1.0000D 06 0.0
NODS NODS19 AREAS 19 1.0000D 06 0.0
NODS NODS20 AREAS 20 1.0000D 06 0.0

```

TRUMP OUTPUT DATA

```

***** MISSLE PROBLEM ***** ONE DIMENSIONAL *****
***** PRINTOUT *****
CYCLE 60          TOO FAST 0          TOO SLOW 0          KMIT 0          DELTIX 12          SMALL 1.00000E 00          TVARY 1.00000E 00          NUTS 7
TOTAL TIME 6.97239E 02          TIME STEP 9.34838E 00          HEAT FLOW -7.26409E 03          FLUX RATE -1.04188E 01          TEMP RATE -7.07565E 00
AVG TEMP 8.62795E 01          HEAT CAPACITY 1.35739D 00          HEAT CONTENT 1.17115E 02          GEN RATE 0.0          HEAT GEN 0.0          TEMP FROM GEN 0.0
*****
NODE 1          TEMP 0.5223D 02          DT 0.375D 00          DD1 522D 01          GEN RATE 0.0          M 1.322E 04          H 0.2945E 05          F 0.2942E 05          CURVE AT 280 F 0.0
      2          0.9177D 02          0.3822D 00          0.3822D 01          0.0          0.3939E 05          0.2396E 01          0.2395E 00
      3          0.9028D 02          0.3722D 00          0.3722D 01          0.0          0.1466E 02          0.4376E 01          0.4367E 01
      4          0.8899D 02          0.3599D 00          0.3599D 01          0.0          0.2056E 02          0.6897E 01          0.6887E 01
      5          0.8522D 02          0.3259D 00          0.3259D 01          0.0          0.2595E 02          0.9839E 01          0.9818E 01
      6          0.8322D 02          0.3125D 00          0.3125D 01          0.0          0.2571E 02          0.9839E 01          0.9818E 01
      7          0.8229D 02          0.3054D 00          0.3054D 01          0.0          0.2535E 02          0.9839E 01          0.9838E 01
      8          0.8176D 02          0.3000D 00          0.3000D 01          0.0          0.2535E 02          0.9839E 01          0.9838E 01
      9          0.8116D 02          0.2968D 00          0.2968D 01          0.0          0.2444E 02          0.9839E 01          0.9838E 01
     10          0.8055D 02          0.2954D 00          0.2954D 01          0.0          0.2444E 02          0.9839E 01          0.9838E 01
     11          0.7993D 02          0.2954D 00          0.2954D 01          0.0          0.2444E 02          0.9839E 01          0.9838E 01
     12          0.7987D 02          0.2954D 00          0.2954D 01          0.0          0.2444E 02          0.9839E 01          0.9838E 01
***** MATERIAL DATA *****
NAME MATL          TOT CAP          TOT HEAT          TGT HEAT          TMELT          HMELT
SAND 1          1.0735E 00          2.3589E 01          2.3589E 01          0.0          0.0
STEL 2          2.8235E 01          8.7174E 01          8.7174E 01          0.0          0.0
AIR 3          1.4424E 03          1.1706E 01          1.1706E 01          0.0          0.0
***** NODE DATA *****
NODE 1          MATL 1          NTYPE 4          RADIUS 0.1000E 02          VOLUME 0.1257E 04          MASS 0.6836D 06          CAPACITY 0.12599E 06          CONDUCTIVITY 0.2700D 03          SLIM 0.0
      2          1          4          0.1500E 01          0.1422E 01          0.1790D 00          0.3477D 01          0.1790D 00          0.1790D 00          0.1790D 00          0.1790D 00
      3          1          4          0.1250E 01          0.1199E 01          0.1577D 00          0.3273D 00          0.1577D 00          0.1577D 00          0.1577D 00          0.1577D 00
      4          1          4          0.1250E 01          0.1277E 01          0.1577D 00          0.3273D 00          0.1577D 00          0.1577D 00          0.1577D 00          0.1577D 00
      5          1          4          0.1250E 01          0.1277E 01          0.1577D 00          0.3273D 00          0.1577D 00          0.1577D 00          0.1577D 00          0.1577D 00
      6          1          4          0.1250E 01          0.1277E 01          0.1577D 00          0.3273D 00          0.1577D 00          0.1577D 00          0.1577D 00          0.1577D 00
      7          1          4          0.1250E 01          0.1277E 01          0.1577D 00          0.3273D 00          0.1577D 00          0.1577D 00          0.1577D 00          0.1577D 00
      8          1          4          0.1250E 01          0.1277E 01          0.1577D 00          0.3273D 00          0.1577D 00          0.1577D 00          0.1577D 00          0.1577D 00
      9          1          4          0.1250E 01          0.1277E 01          0.1577D 00          0.3273D 00          0.1577D 00          0.1577D 00          0.1577D 00          0.1577D 00
     10          1          4          0.1250E 01          0.1277E 01          0.1577D 00          0.3273D 00          0.1577D 00          0.1577D 00          0.1577D 00          0.1577D 00
     11          1          4          0.1250E 01          0.1277E 01          0.1577D 00          0.3273D 00          0.1577D 00          0.1577D 00          0.1577D 00          0.1577D 00
     12          1          4          0.1250E 01          0.1277E 01          0.1577D 00          0.3273D 00          0.1577D 00          0.1577D 00          0.1577D 00          0.1577D 00
***** INTERNAL CONNECTION DATA *****
NODE1 NOD2          AREA 4          D 01          HINT 0E 13          RINT 0.0          TRAN 6D 05          HEAT FLOW 0.0          AVG RATE 0.0
      2          1          2          0.1257D 00          0.1000E 13          0.0          0.0          0.6786D 05          0.4722E 08
      3          1          3          0.1257D 00          0.1000E 13          0.0          0.0          0.6786D 05          0.4722E 08
      4          1          4          0.1257D 00          0.1000E 13          0.0          0.0          0.6786D 05          0.4722E 08
      5          1          5          0.1257D 00          0.1000E 13          0.0          0.0          0.6786D 05          0.4722E 08
      6          1          6          0.1257D 00          0.1000E 13          0.0          0.0          0.6786D 05          0.4722E 08
      7          1          7          0.1257D 00          0.1000E 13          0.0          0.0          0.6786D 05          0.4722E 08
      8          1          8          0.1257D 00          0.1000E 13          0.0          0.0          0.6786D 05          0.4722E 08
      9          1          9          0.1257D 00          0.1000E 13          0.0          0.0          0.6786D 05          0.4722E 08
     10          1          10          0.1257D 00          0.1000E 13          0.0          0.0          0.6786D 05          0.4722E 08
     11          1          11          0.1257D 00          0.1000E 13          0.0          0.0          0.6786D 05          0.4722E 08
     12          1          12          0.1257D 00          0.1000E 13          0.0          0.0          0.6786D 05          0.4722E 08
***** BOUNDARY NODE DATA *****
NODR 7.9876E 01          HEAT FLOW -7.2641D 03          AVG RATE 0.0
2001
SYSTEM TOTAL          -7.2641E 03          -1.9419E 01
***** EXTERNAL CONNECTION DATA *****
NODS NQOSB          AREAS          MSURE 06          POWER 0.0          RSURE 0.0          TRANS 5.6156D 07          HEAT FLOW -7.2641E 03          AVG RATE 0.0
      2          2001          5.6156D 01          1.0000D 06          0.0          0.0          0.0          5.6156D 07          -7.2641E 03          -1.0419E 01
*****

```


PRINTOUT CYCLE 90 TOO FAST TOO SLOW 3 KWT 0 DELTIX 1.5000E 12 SMALL 1.0000E 00 TVARY 1.0000E 00 NUTS 4

TOTAL TIME 8.84236E 02 TIME STEP 7.80307E 00 HEAT FLOW -1.02246E 04 TEMP FROM FLUX -1.15632E 00 FLUX RATE -8.51876E 00 TEMP RATE

AVG TEMP 9.708457E 01 HEAT CAPACITY 3.97390E 00 HEAT CONTENT 1.21448E 52 CGN RATE 0.0 GEN RATE 0.0 HEAT GEN 5.0 TEMP FROM GEN 5.0

TEMP 0.8970 02 DT 0.21000E 01 DDT 9.10E 02 GE N RATE 0.0 W 1.13E 04 H 0.353E 05 F 0.5514E 05 CURE AT 280 F

0.8040 02 0.20970E 01 0.2680E 02 0.0 0.0 0.237E 04 0.387E 05 0.397E 05

0.8620 02 0.21450E 01 0.2800E 02 0.0 0.0 0.237E 04 0.387E 05 0.397E 05

0.9170 02 0.22540E 01 0.2950E 02 0.0 0.0 0.237E 04 0.387E 05 0.397E 05

0.9940 02 0.25980E 01 0.3300E 02 0.0 0.0 0.237E 04 0.387E 05 0.397E 05

0.9770 02 0.27360E 01 0.3450E 02 0.0 0.0 0.237E 04 0.387E 05 0.397E 05

0.1020 03 0.2820E 01 0.3500E 02 0.0 0.0 0.237E 04 0.387E 05 0.397E 05

0.10940 03 0.29160E 01 0.3560E 02 0.0 0.0 0.237E 04 0.387E 05 0.397E 05

MATERIAL DATA

| NAME | MATL | TOT CAP | TOT HEAT | AVG TEMP | TMELT | HMELT |
|------|------|-------------|-------------|------------|-------|-------|
| 2001 | 1 | 2.8237E 01 | 2.67450E 01 | 8.237E 01 | 0.0 | 0.0 |
| 2002 | 2 | 1.44246E 03 | 1.44246E 01 | 1.7033E 02 | 0.0 | 0.0 |

INTERNAL CONNECTION DATA

| NODE | MATL | NTYPE | RADIUS | VOLUME | MASS | CAPACITY | CONDUCTIVITY | ZIP | SLIK |
|------|------|-------|------------|------------|------------|------------|--------------|------------|------------|
| 1 | 1 | 4 | 0.0000E 02 | 0.0000E 04 | 0.0000E 06 | 0.0000E 00 | 0.2700E 03 | 0.0000E 00 | 0.0000E 01 |
| 2 | 2 | 4 | 0.0000E 01 | 0.0000E 04 | 0.0000E 06 | 0.0000E 00 | 0.2700E 03 | 0.0000E 00 | 0.0000E 01 |
| 3 | 3 | 4 | 0.0000E 01 | 0.0000E 04 | 0.0000E 06 | 0.0000E 00 | 0.2700E 03 | 0.0000E 00 | 0.0000E 01 |
| 4 | 4 | 4 | 0.0000E 01 | 0.0000E 04 | 0.0000E 06 | 0.0000E 00 | 0.2700E 03 | 0.0000E 00 | 0.0000E 01 |
| 5 | 5 | 4 | 0.0000E 01 | 0.0000E 04 | 0.0000E 06 | 0.0000E 00 | 0.2700E 03 | 0.0000E 00 | 0.0000E 01 |
| 6 | 6 | 4 | 0.0000E 01 | 0.0000E 04 | 0.0000E 06 | 0.0000E 00 | 0.2700E 03 | 0.0000E 00 | 0.0000E 01 |
| 7 | 7 | 4 | 0.0000E 01 | 0.0000E 04 | 0.0000E 06 | 0.0000E 00 | 0.2700E 03 | 0.0000E 00 | 0.0000E 01 |
| 8 | 8 | 4 | 0.0000E 01 | 0.0000E 04 | 0.0000E 06 | 0.0000E 00 | 0.2700E 03 | 0.0000E 00 | 0.0000E 01 |
| 9 | 9 | 4 | 0.0000E 01 | 0.0000E 04 | 0.0000E 06 | 0.0000E 00 | 0.2700E 03 | 0.0000E 00 | 0.0000E 01 |
| 10 | 10 | 4 | 0.0000E 01 | 0.0000E 04 | 0.0000E 06 | 0.0000E 00 | 0.2700E 03 | 0.0000E 00 | 0.0000E 01 |
| 11 | 11 | 4 | 0.0000E 01 | 0.0000E 04 | 0.0000E 06 | 0.0000E 00 | 0.2700E 03 | 0.0000E 00 | 0.0000E 01 |
| 12 | 12 | 2 | 0.0000E 01 | 0.0000E 04 | 0.0000E 06 | 0.0000E 00 | 0.2700E 03 | 0.0000E 00 | 0.0000E 01 |

BOUNDARY NODE DATA

| NODE | TEMP | HEAT FLOW | AVG RATE |
|------|------------|-------------|-------------|
| 2001 | 1.5937E 02 | -1.0225D 04 | -1.1563E 01 |

SYSTEM TOTAL

| SYSTEM TOTAL | HEAT FLOW | AVG RATE |
|--------------|-------------|-------------|
| 12 | -1.0225E 04 | -1.1563E 01 |

EXTERNAL CONNECTION DATA

| NODS | NODSB | AREAS | HSURE | RSURE | POWER | TRANS | HEAT FLOW | AVG RATE |
|------|-------|------------|------------|-------|-------|------------|-------------|-------------|
| 12 | 2001 | 5.6156D 01 | 1.0000D 06 | 0.0 | 0.0 | 5.6156D 07 | -1.0225E 04 | -1.1563E 01 |

TRUMP OUTPUT DATA

```

***** MISSILE PROBLEM ONE DIMENSIONAL *****
***** PRINTOUT CYCLE 105 *****
***** DELTAX 12 *****
***** SMALL 1.00000E 00 *****
***** TVARY 1.00000E 00 *****
***** NUTS 5 *****
TOTAL TIME 1.07468E 01 HEAT FROM FLUX -1.13916E 01 TEMP RATE -8.59233E 00
1.01237E 05
AVG TEMP 9.89246E 01 HEAT CAPACITY 1.34279E 02 GEN RATE 0.0 HEAT GEN 0.0 TEMP FROM GEN 0.0
***** CURE AT 280 F *****
NODE 1 0.01180 02 0.32570 00 0.03100 01 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00
2 0.00180 02 0.32570 00 0.03100 01 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00
3 0.00830 02 0.44970 00 0.04680 01 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00
4 0.02270 02 0.62730 00 0.07170 01 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00
5 0.08570 02 0.78790 00 0.09150 01 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00
6 0.03340 03 0.91750 00 0.08100 01 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00
7 0.01140 03 0.92850 00 0.08830 01 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00
8 0.00970 03 0.94460 00 0.88830 01 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00
9 0.01600 03 0.97460 00 0.90840 01 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00
10 0.012380 03 0.97630 00 0.91670 01 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00
11 0.012380 03 0.98510 00 0.91670 01 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00
12 0.012380 03 0.98510 00 0.91670 01 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00 0.00000 00
***** MATERIAL DATA *****
NAME MATL TOI CAP TOI HEAT TOI TWELT HMETL
SAND 1 1.07359E 00 1.0417E 02 9.7038E 01 0.0 0.0
STEL 2 2.82356E 01 2.9940E 01 1.0603E 02 0.0 0.0
AIR 3 1.44246E 03 1.6749E 01 1.1612E 02 0.0 0.0
***** MODE DATA *****
MODE MATL NTYPE VOLUME 04 MASS 06 CAPACITY 06 CONDUCTIVITY 06 ZIP 05 SLIM 05
1 1 1 4 0.342E 01 0.17090 00 0.32700 01 0.2700E 03 0.67800 02 0.19140 01
2 1 1 4 0.924E 01 0.45270 00 0.7000E 03 0.7000E 03 0.50000 02 0.00000 00 0.00000 00
3 1 1 4 0.1500E 01 0.07370 00 0.0000E 03 0.0000E 03 0.00000 02 0.00000 00 0.00000 00
4 1 1 4 0.227E 01 0.11380 00 0.0000E 03 0.0000E 03 0.00000 02 0.00000 00 0.00000 00
5 1 1 4 0.5237E 01 0.26280 01 0.22800 01 0.26400 01 0.35270 01 0.00000 00 0.00000 00
6 1 1 4 0.5500E 01 0.28400 02 0.44000 02 0.44000 02 0.52000 02 0.00000 00 0.00000 00
7 1 1 4 0.7500E 01 0.40120 03 0.49300 03 0.49300 03 0.56000 03 0.00000 00 0.00000 00
8 1 1 4 0.8468E 01 0.47800 03 0.57000 03 0.57000 03 0.64000 03 0.00000 00 0.00000 00
9 1 1 4 0.9738E 01 0.54700 03 0.65000 03 0.65000 03 0.72000 03 0.00000 00 0.00000 00
10 10 2 2 0.1000E 02 0.1000E 02 0.1000E 02 0.1000E 02 0.1000E 02 0.00000 00 0.00000 00
11 11 2 2 0.1000E 02 0.1000E 02 0.1000E 02 0.1000E 02 0.1000E 02 0.00000 00 0.00000 00
12 12 2 2 0.1000E 02 0.1000E 02 0.1000E 02 0.1000E 02 0.1000E 02 0.00000 00 0.00000 00
***** INTERNAL CONNECTION DATA *****
NOD1 NOD2 AREA 01 HINT 04 RINT 04 TRAN 05 AVG RATE 09
1 2 0.62800 01 0.1000E 13 0.0 0.67800 05 0.32700 01
2 3 0.12800 01 0.1000E 13 0.0 0.32700 01 0.32700 01
3 4 0.12800 01 0.1000E 13 0.0 0.32700 01 0.32700 01
4 5 0.35140 02 0.1000E 13 0.0 0.67800 05 0.32700 01
5 7 0.36700 02 0.1000E 13 0.0 0.67800 05 0.32700 01
6 8 0.37700 02 0.1000E 13 0.0 0.67800 05 0.32700 01
7 9 0.50500 02 0.1000E 13 0.0 0.67800 05 0.32700 01
8 9 0.50500 02 0.1000E 13 0.0 0.67800 05 0.32700 01
9 10 0.50500 02 0.1000E 13 0.0 0.67800 05 0.32700 01
10 11 0.50500 02 0.1000E 13 0.0 0.67800 05 0.32700 01
11 12 0.50500 02 0.1000E 13 0.0 0.67800 05 0.32700 01
***** BOUNDARY NODE DATA *****
NODR TEMPR 02 HEAT FLOW 04 AVG RATE 01
2001 1.2380E 02 -1.15330 04 -1.1392E 01
***** SYSTEM TOTAL *****
SYSTEM TOTAL -1.1533E 04 -1.1392E 01
***** EXTERNAL CONNECTION DATA *****
NODS NODSB AREAS HSURE 06 RSURE 06 TRANS 07 HEAT FLOW 04 AVG RATE 01
12 2001 5.61560 01 1.00000 06 0.0 5.41560 07 -1.1533E 04 -1.1392E 01

```


TRUMP OUTPUT DATA

```

*****
* MISSLE PROBLEM ONE DIMENSIONAL
*****
PRINTOUT CYCLE 120          TOD SLOW 0          KMIT 0          MELTIX 12          SMALL 1.000000 00          TIVARY 1.000000 00          NUTS 5
TOTAL TIME 1.8970E 05      TIME STEP 1.3413E 01      HEAT FLOW -1.82268E 04      TEMP FROM FLUX -7.2863E 03      FLUX RATE -1.05289E 01      TEMP RATE -7.75673E 00
AVG TEMP 1.11428E 02      HEAT CAPACITY 1.35739D 00      HEAT CONTENT 1.51250E 02      GEN RATE 0.0          HEAT GEN 0.0          TEMP FROM GEN 0.0
*****
NODE 01 0.9512D 02      DT 0.8219D 00      DDT 0.6118D 01      GE N RATE 0.1287E 04      H 0.2895E 05      F 0.1971E 05      CURE AT 280 F
01003D 03      0.8429D 00      0.6218D 00      0.0000 00      0.3219E 01      0.2247E 01      0.5244E 00
01007D 03      0.9271D 00      0.6521D 00      0.0000 00      0.7714E 01      0.2247E 01      0.5244E 00
01015D 03      0.9676D 00      0.6824D 00      0.0000 00      0.9099E 01      0.2247E 01      0.5244E 00
01113D 03      0.9884D 00      0.7136D 01      0.0000 00      0.0320E 02      0.1203E 01      0.1203E 01
01127D 03      0.9886D 00      0.7136D 01      0.0000 00      0.0320E 02      0.1203E 01      0.1203E 01
01127D 03      0.9886D 00      0.7136D 01      0.0000 00      0.0320E 02      0.1203E 01      0.1203E 01
01130D 03      0.9886D 00      0.7136D 01      0.0000 00      0.0320E 02      0.1203E 01      0.1203E 01
01133D 03      0.8922D 00      0.6667D 01      0.0000 00      0.4201E 02      0.1055E 02      0.1055E 02
*****
MATERIAL DATA
NAME MATL TOT CAP TOT HEAT TOT HEAT H MELT
SAND 1 1.0735E 00 1.1142E 02 1.0911E 02 0.0 0.0
STEEL 2 2.8235E 01 3.3199E 01 1.2013E 02 0.0 0.0
AIR 3 1.4424E 03 1.8736E 01 1.2992E 02 0.0 0.0
*****
NODE DATA
NODE MATL NTYPE VOLUME 04 MASS 00 CAPACITY 03 CONDUCTIVITY 05 ZIP 860 05 SLIM 00
1 1 1 4 0.3127E 01 0.6830D 00 0.1299D 01 0.2700D 03 0.4786D 05 0.9100D 00 0.0000 00
2 1 1 4 0.1170D 00 0.1700D 00 0.2700D 03 0.2700D 03 0.4786D 05 0.9100D 00 0.0000 00
3 1 1 4 0.1874D 00 0.2700D 03 0.2700D 03 0.2700D 03 0.4786D 05 0.9100D 00 0.0000 00
4 1 1 4 0.1153D 01 0.1700D 00 0.2700D 03 0.2700D 03 0.4786D 05 0.9100D 00 0.0000 00
5 1 1 4 0.2287E 02 0.1378D 00 0.2700D 03 0.2700D 03 0.4786D 05 0.9100D 00 0.0000 00
6 1 1 4 0.2287E 02 0.1378D 00 0.2700D 03 0.2700D 03 0.4786D 05 0.9100D 00 0.0000 00
7 1 1 4 0.2287E 02 0.1378D 00 0.2700D 03 0.2700D 03 0.4786D 05 0.9100D 00 0.0000 00
8 1 1 4 0.2287E 02 0.1378D 00 0.2700D 03 0.2700D 03 0.4786D 05 0.9100D 00 0.0000 00
9 1 1 4 0.2287E 02 0.1378D 00 0.2700D 03 0.2700D 03 0.4786D 05 0.9100D 00 0.0000 00
10 1 1 4 0.2287E 02 0.1378D 00 0.2700D 03 0.2700D 03 0.4786D 05 0.9100D 00 0.0000 00
11 1 1 4 0.2287E 02 0.1378D 00 0.2700D 03 0.2700D 03 0.4786D 05 0.9100D 00 0.0000 00
12 1 1 4 0.2287E 02 0.1378D 00 0.2700D 03 0.2700D 03 0.4786D 05 0.9100D 00 0.0000 00
*****
INTERNAL CONNECTION DATA
NODE1 NOD2 AREA 01 HINT 04 RINT 05 TRAN 60D 05 HEAT FLOW 05 AVG RATE 08
001 2 3 0.2287D 01 0.1000E 13 0.0000 00 0.6788D 05 0.1871E 08
001 3 4 0.1299D 01 0.1000E 13 0.0000 00 0.4698D 05 0.4698E 08
001 4 5 0.1874D 01 0.1000E 13 0.0000 00 0.3219D 05 0.3219E 08
001 5 6 0.1153D 01 0.1000E 13 0.0000 00 0.7714D 05 0.7714E 08
001 6 7 0.2287D 01 0.1000E 13 0.0000 00 0.9886D 05 0.9886E 08
001 7 8 0.2287D 01 0.1000E 13 0.0000 00 0.9886D 05 0.9886E 08
001 8 9 0.2287D 01 0.1000E 13 0.0000 00 0.9886D 05 0.9886E 08
001 9 10 0.2287D 01 0.1000E 13 0.0000 00 0.9886D 05 0.9886E 08
001 10 11 0.2287D 01 0.1000E 13 0.0000 00 0.9886D 05 0.9886E 08
001 11 12 0.2287D 01 0.1000E 13 0.0000 00 0.9886D 05 0.9886E 08
*****
BOUNDARY NODE DATA
NOD0 TEMP 1.5759E 02 HEAT FLOW 04 AVG RATE 01
2001 -1.2527D 04 -1.0529E 01
*****
SYSTEM TOTAL -1.2527E 04 -1.0529E 01
EXTERNAL CONNECTION DATA
NODS NODSB AREAS 01 HSURE 06 POWER 3.0          RSURE 5.6156D 07 HEAT FLOW 07 AVG RATE
12 2001 5.6156D 01 1.0300D 06 0.0          3.0          5.6156D 07 1.321E 04 1.321E 04
WILL REPEAT CYCLE 130          DTMAX = 2.865E 00          DTRE = 1.720E 00          DELT = 1.552D 01          SUMTH = 1.321E 03
WILL REPEAT CYCLE 130          DTMAX = 2.865E 00          DTRE = 1.720E 00          DELT = 1.552D 01          SUMTH = 1.321E 03
*****

```



```

//WIR71687 JOB (1687,0860FT,NF12),WIRZBURGER.A. BOX W',TIME=(4,00)
//JOBLIB DD UNIT=2321,DSNAME=SI734.KATZ,
//DISP=(OLD,PASS),VOLUME=SER=CELOO1
//EXEC PGM=TRUMP,REGION=350K
//FT06F001 DD SYSOUT=A,DCB=(RECFM=FBA,LRECL=133,BLKSIZE=3325),
//FT05F001 DD SPACE=(CYL,(6,1))
//FT05F001 DD *

```

* MISSILE PROBLEM TWO DIMENSIONAL

```

BLOCK 1 CONTROLS, LIMITS, CONSTANTS
5 1 1.000 E 00 1.000E 00 1380.0 8000.0
114.0

```

```

BLOCK 2 MATERIAL NAMES, NUMBERS, THERMAL PROPERTIES.
ASAND 1 0.0544 0.19 0.00027
ASTEL 2 0.2807 0.109 0.0364
AAIR 3 0.0000436 0.240 0.0000225

```

```

BLOCK 4 NODE NUMBERS, MATERIAL REFERENCES, TYPES, VOLUMES.
1 1 0.25 0.002 0.001
2 1 0.25 0.998 0.501
3 1 0.25 1.00 1.50
4 1 0.25 1.00 3.50
5 1 0.25 1.00 4.50
6 1 0.25 1.75 5.375
7 1 0.25 1.00 5.875
8 1 0.25 1.00 6.50
9 1 0.25 1.00 7.50
10 1 0.25 0.9375 8.46875
11 1 0.0 0.0 8.9375
12 1 0.0 0.0 0.0

```

```

BLOCK 5 INTERNAL THERMAL CONNECTION NODE NUMBERS.
1 2 0.001 0.499 0.002
2 12 0.499 0.500 1.00
3 3 12 0.500 0.25 1.00
4 3 12 0.500 0.25 3.00
5 3 12 0.500 0.25 4.00
6 3 12 0.375 0.25 5.00
7 3 12 0.125 0.25 5.75
8 3 12 0.500 0.25 6.00
9 3 12 0.125 0.25 6.00
10 1 12 0.500 0.25 7.00
11 1 12 0.46875 0.25 8.00
12 1 12 0.000785 0.25 8.9375
13 1 12 0.000785 0.000785 0.001273

```

0.00008478

| | | | | | |
|----|-----|---------|---------|-------|------------|
| 23 | 38 | 0.393 | 0.393 | 0.637 | 0.00000729 |
| 34 | 39 | 1.178 | 1.178 | 0.637 | 0.00000729 |
| 45 | 40 | 1.748 | 1.748 | 0.637 | 0.00000729 |
| 56 | 41 | 2.534 | 2.534 | 0.637 | 0.00000729 |
| 67 | 42 | 3.271 | 3.271 | 0.637 | 0.00000729 |
| 78 | 43 | 4.105 | 4.105 | 0.559 | 0.00000729 |
| 89 | 44 | 5.105 | 5.105 | 0.637 | 0.00000729 |
| 10 | 45 | 6.651 | 6.651 | 0.637 | 0.00000729 |
| 11 | 46 | 8.937 | 8.937 | 0.637 | 0.00000729 |
| 18 | 47 | 10.000 | 10.000 | 0.637 | 0.00000729 |
| 20 | 48 | 11.000 | 11.000 | 0.637 | 0.00000729 |
| 24 | 49 | 12.255 | 12.255 | 0.637 | 0.00000729 |
| 32 | 50 | 13.637 | 13.637 | 0.637 | 0.00000729 |
| 44 | 51 | 15.178 | 15.178 | 0.637 | 0.00000729 |
| 12 | 52 | 16.748 | 16.748 | 0.637 | 0.00000729 |
| 13 | 53 | 18.534 | 18.534 | 0.637 | 0.00000729 |
| 14 | 54 | 21.000 | 21.000 | 0.637 | 0.00000729 |
| 15 | 55 | 23.255 | 23.255 | 0.637 | 0.00000729 |
| 16 | 56 | 25.500 | 25.500 | 0.637 | 0.00000729 |
| 17 | 57 | 27.755 | 27.755 | 0.637 | 0.00000729 |
| 18 | 58 | 30.000 | 30.000 | 0.637 | 0.00000729 |
| 3 | 59 | 32.255 | 32.255 | 0.637 | 0.00000729 |
| 4 | 60 | 34.500 | 34.500 | 0.637 | 0.00000729 |
| 5 | 61 | 36.755 | 36.755 | 0.637 | 0.00000729 |
| 6 | 62 | 39.000 | 39.000 | 0.637 | 0.00000729 |
| 7 | 63 | 41.255 | 41.255 | 0.637 | 0.00000729 |
| 8 | 64 | 43.500 | 43.500 | 0.637 | 0.00000729 |
| 9 | 65 | 45.755 | 45.755 | 0.637 | 0.00000729 |
| 10 | 66 | 48.000 | 48.000 | 0.637 | 0.00000729 |
| 11 | 67 | 50.255 | 50.255 | 0.637 | 0.00000729 |
| 11 | 68 | 52.500 | 52.500 | 0.637 | 0.00000729 |
| 11 | 69 | 54.755 | 54.755 | 0.637 | 0.00000729 |
| 11 | 70 | 57.000 | 57.000 | 0.637 | 0.00000729 |
| 11 | 71 | 59.255 | 59.255 | 0.637 | 0.00000729 |
| 11 | 72 | 61.500 | 61.500 | 0.637 | 0.00000729 |
| 11 | 73 | 63.755 | 63.755 | 0.637 | 0.00000729 |
| 11 | 74 | 66.000 | 66.000 | 0.637 | 0.00000729 |
| 11 | 75 | 68.255 | 68.255 | 0.637 | 0.00000729 |
| 11 | 76 | 70.500 | 70.500 | 0.637 | 0.00000729 |
| 11 | 77 | 72.755 | 72.755 | 0.637 | 0.00000729 |
| 11 | 78 | 75.000 | 75.000 | 0.637 | 0.00000729 |
| 11 | 79 | 77.255 | 77.255 | 0.637 | 0.00000729 |
| 11 | 80 | 79.500 | 79.500 | 0.637 | 0.00000729 |
| 11 | 81 | 81.755 | 81.755 | 0.637 | 0.00000729 |
| 11 | 82 | 84.000 | 84.000 | 0.637 | 0.00000729 |
| 11 | 83 | 86.255 | 86.255 | 0.637 | 0.00000729 |
| 11 | 84 | 88.500 | 88.500 | 0.637 | 0.00000729 |
| 11 | 85 | 90.755 | 90.755 | 0.637 | 0.00000729 |
| 11 | 86 | 93.000 | 93.000 | 0.637 | 0.00000729 |
| 11 | 87 | 95.255 | 95.255 | 0.637 | 0.00000729 |
| 11 | 88 | 97.500 | 97.500 | 0.637 | 0.00000729 |
| 11 | 89 | 99.755 | 99.755 | 0.637 | 0.00000729 |
| 11 | 90 | 102.000 | 102.000 | 0.637 | 0.00000729 |
| 11 | 91 | 104.255 | 104.255 | 0.637 | 0.00000729 |
| 11 | 92 | 106.500 | 106.500 | 0.637 | 0.00000729 |
| 11 | 93 | 108.755 | 108.755 | 0.637 | 0.00000729 |
| 11 | 94 | 111.000 | 111.000 | 0.637 | 0.00000729 |
| 11 | 95 | 113.255 | 113.255 | 0.637 | 0.00000729 |
| 11 | 96 | 115.500 | 115.500 | 0.637 | 0.00000729 |
| 11 | 97 | 117.755 | 117.755 | 0.637 | 0.00000729 |
| 11 | 98 | 120.000 | 120.000 | 0.637 | 0.00000729 |
| 11 | 99 | 122.255 | 122.255 | 0.637 | 0.00000729 |
| 11 | 100 | 124.500 | 124.500 | 0.637 | 0.00000729 |

| | | | | | | | |
|---------|----------|-------------|-----------|--------|--------|--------|----|
| BLOCK 6 | EXTERNAL | 12 | 1 | 0.25 | 8.9375 | 1.00 | E6 |
| BLOCK 7 | BOUNDARY | TEMPERATURE | VARIATION | 1440.0 | 46.5 | 1380.0 | |
| 2001 | 100 | 109.5 | 1440.0 | 43.0 | 1284.0 | | |
| 2002 | 100 | 109.0 | 1440.0 | 25.0 | 1212.0 | | |
| 2003 | 100 | 96.0 | 1440.0 | 29.5 | 1284.0 | | |
| 2004 | 100 | 102.5 | 1440.0 | | | | |

ENDED-1 LAST CARD OF DATA DECK

MISSILE PROBLEM TWO DIMENSIONAL
 INPUT UNIT = 5. OUTPUT UNIT = 6.
 DATA BLOCK 10 CONTROLS, LIMITS, CONSTANTS

IPRINT NUM KDATA KSPEC MCYC MSEC NPUNCH NDOT IRITE SCALE
 0 0 0 30000 30000 0 0 0.10000E 01
 KC2 KT3 DELT0 SMALL TVARY TMAX TMIN TMAX TMIN
 1.00000E 12 1.00000E 00 1.00000E 00 0.0 1.4000E 03 -1.00000E 12 1.00000E 12
 KD KSYM GEOM SIGMA TBASE
 2 1 6.283190 00 1.73000E-09 4.60000E 02

TOVE 02 0.0 BONE 0.0 FONE 0.0 HONE 0.0 PONE 0.0
 1.00000E 00 0.00000E 00 0.00000E 00 0.00000E 00 0.00000E 00 0.00000E 00
 DATA BLOCK 20 MATERIAL NAMES, NUMBERS, THE MAL PROPERTIES.

NAME MATL : INDEX KA KB LTABC LTABX DENSITY CAPACITY CONDUCTIVITY TMELT HMELT
 1 SAND 1 0 0 0 5.400E-02 1.0000E-01 2.4000E-01 0.0 0.0
 2 AIR 2 0 0 0 2.500E-05 2.4000E-01 2.2500E-05 0.0 0.0
 DATA BLOCK 40 NODE NUMBERS, MATERIAL REFERENCES, TYPES, VOLUMES.

| NODE | INDEX | MATL | NTYPE | DLONG | DWIDE | DRAD | VOLUME |
|------|-------|------|-------|-------------|-------------|-------------|-------------|
| 1 | 1 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 2 | 2 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 3 | 3 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 4 | 4 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 5 | 5 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 6 | 6 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 7 | 7 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 8 | 8 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 9 | 9 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 10 | 10 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 11 | 11 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 12 | 12 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 13 | 13 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 14 | 14 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 15 | 15 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 16 | 16 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 17 | 17 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 18 | 18 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 19 | 19 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 20 | 20 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 21 | 21 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 22 | 22 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 23 | 23 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 24 | 24 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 25 | 25 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 26 | 26 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 27 | 27 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 28 | 28 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 29 | 29 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 30 | 30 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 31 | 31 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 32 | 32 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 33 | 33 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 34 | 34 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 35 | 35 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 36 | 36 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 37 | 37 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 38 | 38 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 39 | 39 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |
| 40 | 40 | 1 | 0 | 0.00000E 01 | 0.00000E 03 | 1.00000E 00 | 3.14159E 06 |

DATA BLOCK 50 INTERNAL THERMAL CONNECTION NODE NUMBERS

| NCD1 | NOD2 | INDEX | DEL1 | DLONG | DRAD | HINT | RINT | AREA |
|------|------|-------|------|-------|------|------|------|------|
| 1 | 13 | 1 | 1 | 50 | 00 | 00 | 00 | 14 |
| 2 | 26 | 2 | 1 | 50 | 00 | 00 | 00 | 14 |
| 3 | 26 | 3 | 1 | 50 | 00 | 00 | 00 | 14 |
| 4 | 26 | 4 | 1 | 50 | 00 | 00 | 00 | 14 |
| 5 | 26 | 5 | 1 | 50 | 00 | 00 | 00 | 14 |
| 6 | 26 | 6 | 1 | 50 | 00 | 00 | 00 | 14 |
| 7 | 26 | 7 | 1 | 50 | 00 | 00 | 00 | 14 |
| 8 | 26 | 8 | 1 | 50 | 00 | 00 | 00 | 14 |
| 9 | 26 | 9 | 1 | 50 | 00 | 00 | 00 | 14 |
| 10 | 26 | 10 | 1 | 50 | 00 | 00 | 00 | 14 |
| 11 | 26 | 11 | 1 | 50 | 00 | 00 | 00 | 14 |
| 12 | 26 | 12 | 1 | 50 | 00 | 00 | 00 | 14 |
| 13 | 26 | 13 | 1 | 50 | 00 | 00 | 00 | 14 |
| 14 | 26 | 14 | 1 | 50 | 00 | 00 | 00 | 14 |
| 15 | 26 | 15 | 1 | 50 | 00 | 00 | 00 | 14 |
| 16 | 26 | 16 | 1 | 50 | 00 | 00 | 00 | 14 |
| 17 | 26 | 17 | 1 | 50 | 00 | 00 | 00 | 14 |
| 18 | 26 | 18 | 1 | 50 | 00 | 00 | 00 | 14 |
| 19 | 26 | 19 | 1 | 50 | 00 | 00 | 00 | 14 |
| 20 | 26 | 20 | 1 | 50 | 00 | 00 | 00 | 14 |
| 21 | 26 | 21 | 1 | 50 | 00 | 00 | 00 | 14 |
| 22 | 26 | 22 | 1 | 50 | 00 | 00 | 00 | 14 |
| 23 | 26 | 23 | 1 | 50 | 00 | 00 | 00 | 14 |
| 24 | 26 | 24 | 1 | 50 | 00 | 00 | 00 | 14 |
| 25 | 26 | 25 | 1 | 50 | 00 | 00 | 00 | 14 |
| 26 | 26 | 26 | 1 | 50 | 00 | 00 | 00 | 14 |
| 27 | 26 | 27 | 1 | 50 | 00 | 00 | 00 | 14 |
| 28 | 26 | 28 | 1 | 50 | 00 | 00 | 00 | 14 |
| 29 | 26 | 29 | 1 | 50 | 00 | 00 | 00 | 14 |
| 30 | 26 | 30 | 1 | 50 | 00 | 00 | 00 | 14 |
| 31 | 26 | 31 | 1 | 50 | 00 | 00 | 00 | 14 |
| 32 | 26 | 32 | 1 | 50 | 00 | 00 | 00 | 14 |
| 33 | 26 | 33 | 1 | 50 | 00 | 00 | 00 | 14 |
| 34 | 26 | 34 | 1 | 50 | 00 | 00 | 00 | 14 |
| 35 | 26 | 35 | 1 | 50 | 00 | 00 | 00 | 14 |
| 36 | 26 | 36 | 1 | 50 | 00 | 00 | 00 | 14 |
| 37 | 26 | 37 | 1 | 50 | 00 | 00 | 00 | 14 |
| 38 | 26 | 38 | 1 | 50 | 00 | 00 | 00 | 14 |
| 39 | 26 | 39 | 1 | 50 | 00 | 00 | 00 | 14 |
| 40 | 26 | 40 | 1 | 50 | 00 | 00 | 00 | 14 |
| 41 | 26 | 41 | 1 | 50 | 00 | 00 | 00 | 14 |
| 42 | 26 | 42 | 1 | 50 | 00 | 00 | 00 | 14 |
| 43 | 26 | 43 | 1 | 50 | 00 | 00 | 00 | 14 |
| 44 | 26 | 44 | 1 | 50 | 00 | 00 | 00 | 14 |
| 45 | 26 | 45 | 1 | 50 | 00 | 00 | 00 | 14 |
| 46 | 26 | 46 | 1 | 50 | 00 | 00 | 00 | 14 |
| 47 | 26 | 47 | 1 | 50 | 00 | 00 | 00 | 14 |
| 48 | 26 | 48 | 1 | 50 | 00 | 00 | 00 | 14 |
| 49 | 26 | 49 | 1 | 50 | 00 | 00 | 00 | 14 |
| 50 | 26 | 50 | 1 | 50 | 00 | 00 | 00 | 14 |

| MOD1 | MOD2 | INDEX | DEL1 | DEL2 | C | LONG | GRAD | HINT | RINT | AREA |
|------|------|-------|-------------|-------------|---|-------|------|------|------|-------------|
| 10 | 46 | 58 | 5.89105D 00 | 5.8905D 00 | 2 | 50000 | 00 | 01 | 00 | 1.3777D 00 |
| 11 | 47 | 59 | 6.65100D 00 | 6.65100D 00 | 2 | 50000 | 00 | 01 | 00 | 1.42448D 00 |
| 320 | 24 | 61 | 6.22500D 02 | 6.22500D 02 | 2 | 50000 | 00 | 01 | 00 | 1.42448D 00 |
| 44 | 12 | 62 | 6.22500D 02 | 6.22500D 02 | 2 | 50000 | 00 | 01 | 00 | 1.42448D 00 |
| 44 | 13 | 63 | 6.22500D 02 | 6.22500D 02 | 2 | 50000 | 00 | 01 | 00 | 1.42448D 00 |
| 13 | 25 | 64 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 24 | 35 | 65 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 35 | 12 | 66 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 14 | 67 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 24 | 15 | 68 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 35 | 17 | 69 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 15 | 70 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 71 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 72 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 73 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 74 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 75 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 76 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 77 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 78 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 79 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 80 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 81 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 82 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 83 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 84 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 85 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 86 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 87 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 88 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 89 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 90 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 91 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 92 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 93 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 94 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 95 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 96 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 97 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 98 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 99 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |
| 12 | 17 | 100 | 7.85000D 04 | 7.85000D 04 | 2 | 50000 | 00 | 01 | 00 | 1.99966D 03 |

| DATA BLOCK | NODS | NODSB | INDEX | LTARH | POWER | EXTERNAL | THERMAL CONNECTIONS | DRAD | HSURE | PSURE | AREAS |
|------------|------|-------|-------|-------|-------|------------|---------------------|------------|-------|------------|-------|
| 12 | 2001 | 1 | 1 | 0 | 0.0 | 8.9375E 00 | 06 | 1.0000D 06 | 0.0 | 1.4039D 01 | |
| 24 | 2002 | 2 | 2 | 0 | 0.0 | 8.9375E 00 | 06 | 1.0000D 06 | 0.0 | 1.4039D 01 | |
| 36 | 2003 | 3 | 3 | 0 | 0.0 | 8.9375E 00 | 06 | 1.0000D 06 | 0.0 | 1.4039D 01 | |
| 48 | 2004 | 4 | 4 | 0 | 0.0 | 8.9375E 00 | 06 | 1.0000D 06 | 0.0 | 1.4039D 01 | |

DATA BLOCK 70 BOUNDARY TEMPERATURE VARIATION.

NODB INDEX 1
 2001

| TEMPB | SLOPE | TIMEB |
|--------------|----------------|--------------|
| 8.400000E 01 | -4.166666E -02 | 1.200000E 02 |
| 7.900000E 01 | -5.833333E -02 | 1.400000E 02 |
| 7.100000E 01 | -8.833333E -02 | 3.600000E 02 |
| 6.400000E 01 | -1.833333E -01 | 4.800000E 02 |
| 6.800000E 01 | 1.666666E -01 | 7.200000E 02 |
| 1.400000E 02 | 3.083333E -01 | 8.400000E 02 |
| 1.600000E 02 | 2.283333E -01 | 9.600000E 02 |
| 1.500000E 02 | -2.600000E -01 | 1.080000E 03 |
| 1.300000E 02 | -1.266666E -01 | 1.200000E 03 |
| 1.900000E 01 | -2.666666E -01 | 1.440000E 03 |

NODB INDEX 2
 2002

| TEMPB | SLOPE | TIMEB |
|--------------|----------------|--------------|
| 8.700000E 01 | -4.166666E -02 | 1.200000E 02 |
| 8.200000E 01 | -4.333333E -02 | 1.400000E 02 |
| 7.300000E 01 | -3.300000E -02 | 3.600000E 02 |
| 7.700000E 01 | -8.000000E -02 | 4.800000E 02 |
| 7.900000E 01 | 3.583333E -01 | 7.200000E 02 |
| 1.400000E 02 | 1.416666E -01 | 8.400000E 02 |
| 1.300000E 02 | 1.966666E -01 | 9.600000E 02 |
| 1.470000E 02 | -1.833333E -01 | 1.080000E 03 |
| 1.030000E 02 | -3.666666E -01 | 1.200000E 03 |

NODB INDEX 3
 2003

| TEMPB | SLOPE | TIMEB |
|--------------|----------------|--------------|
| 9.100000E 01 | -3.333333E -02 | 1.200000E 02 |
| 8.700000E 01 | -3.333333E -02 | 1.400000E 02 |
| 8.500000E 01 | -3.333333E -02 | 3.600000E 02 |
| 7.900000E 01 | -3.333333E -02 | 4.800000E 02 |
| 7.800000E 01 | 5.000000E -01 | 7.200000E 02 |
| 9.800000E 01 | 1.666666E -01 | 8.400000E 02 |
| 1.060000E 02 | 1.666666E -01 | 9.600000E 02 |
| 1.180000E 02 | 4.166666E -01 | 1.080000E 03 |
| 1.210000E 02 | 5.833333E -01 | 1.200000E 03 |
| 1.030000E 02 | -1.500000E -01 | 1.440000E 03 |

NODB INDEX 4
 2004

| TEMPB | SLOPE | TIMEB |
|--------------|----------------|--------------|
| 9.600000E 01 | 5.000000E -02 | 1.200000E 02 |
| 9.000000E 01 | -4.166666E -02 | 1.400000E 02 |
| 8.500000E 01 | -3.333333E -02 | 3.600000E 02 |
| 7.500000E 01 | -7.000000E -02 | 4.800000E 02 |
| 7.800000E 01 | 7.999999E -02 | 7.200000E 02 |
| 1.060000E 02 | 1.333333E -01 | 8.400000E 02 |
| 1.110000E 02 | 9.166666E -02 | 9.600000E 02 |
| 1.220000E 02 | 9.166666E -02 | 1.080000E 03 |
| 1.310000E 02 | 7.499999E -02 | 1.200000E 03 |
| 1.300000E 02 | -8.333333E -03 | 1.320000E 03 |
| 1.090000E 02 | -1.750000E -01 | 1.440000E 03 |

DATA ENDED -10
 LAST CARD OF DATA DECK

DATA DECK 1

| TRUMP OUTPUT DATA | | TWO DIMENSIONAL | | | | | | | | | | MATERIAL DATA | | |
|--|----------|-----------------|-----|-----------|-----|-----|-----|---------------|-----|-----|-----|---------------|------|------|
| * MISSILE PROBLEM | | | | | | | | | | | | MATERIAL DATA | | |
| PRINTOUT CYCLE TOG FAST TOO SLOW KWT DELT MXE-05 SMALL TVARY NUTS | | | | | | | | | | | | MATERIAL DATA | | |
| TOTAL TIME 1.00000E-12 TIME STEP 1.00000E-12 HEAT FLOW 4.29983E-14 SMP FROM FLUX 3.16773E-02 FLUX RATE 4.29984E-02 TEMP RATE 3.16773E-02 | | | | | | | | | | | | MATERIAL DATA | | |
| AVG TEMP 1.4000E 02 HEAT CAPACITY 1.35739D 00 HEAT CONTENT 1.54742E 02 GEN RATE 0.0 HEAT GEN 0.0 TEMP FROM GEN 0.0 | | | | | | | | | | | | MATERIAL DATA | | |
| NODE | TEMP | DT | DDT | GE N RATE | W | H | F | CURE AT 280 F | W | H | F | CURE AT 280 F | NAME | MATL |
| 1 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 1 | STEL |
| 2 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 2 | STEL |
| 3 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 3 | AIR |
| 4 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 4 | AIR |
| 5 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 5 | AIR |
| 6 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 6 | AIR |
| 7 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 7 | AIR |
| 8 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 8 | AIR |
| 9 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 9 | AIR |
| 10 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 10 | AIR |
| 11 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 11 | AIR |
| 12 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 12 | AIR |
| 13 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 13 | AIR |
| 14 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 14 | AIR |
| 15 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 15 | AIR |
| 16 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 16 | AIR |
| 17 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 17 | AIR |
| 18 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 18 | AIR |
| 19 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 19 | AIR |
| 20 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 20 | AIR |
| 21 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 21 | AIR |
| 22 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 22 | AIR |
| 23 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 23 | AIR |
| 24 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 24 | AIR |
| 25 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 25 | AIR |
| 26 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 26 | AIR |
| 27 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 27 | AIR |
| 28 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 28 | AIR |
| 29 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 29 | AIR |
| 30 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 30 | AIR |
| 31 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 31 | AIR |
| 32 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 32 | AIR |
| 33 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 33 | AIR |
| 34 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 34 | AIR |
| 35 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 35 | AIR |
| 36 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 36 | AIR |
| 37 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 37 | AIR |
| 38 | 11400 03 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 | 38 | AIR |

TRUMP OUTPUT DATA

* MISSILE PROBLEM TWO DIMENSIONAL

DATA DECK 1

PRINTOUT CYCLE 40 TIME STEP 11 DELTMX 12 SMALL 1.00000E 00 TVARY 1.00000E 00 NUTS 6

TOTAL TIME 3.07043E 02 TIME STEP 1.66156E 01 HEAT FLOW 3.59496E 03 FLUX RATE 1.17083E 01 TEMP RATE 8.62564E 00

AVG TEMP 9.94270E 01 HEAT CAPACITY 1.35739D 00 GEN RATE 0.0 HEAT GEN 0.0 TEMP FROM GEN 0.0

| NODE | TEMP | DT | DT CAP | TOT CAP | TOT HEAT | AVG TEMP | T MELT | H MELT | CURE AT 280 F |
|------|-------|---------|---------|---------|----------|----------|---------|---------|---------------|
| 1 | 11000 | 0.26400 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 2 | 11000 | 0.62500 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 3 | 11000 | 0.62500 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 4 | 10770 | 0.66650 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 5 | 10770 | 0.65550 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 6 | 10690 | 0.68800 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 7 | 10690 | 0.68300 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 8 | 10420 | 0.65500 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 9 | 10420 | 0.71200 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 10 | 10500 | 0.63500 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 11 | 10500 | 0.73400 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 12 | 10100 | 0.71700 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 13 | 10360 | 0.69300 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 14 | 98000 | 0.72200 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 15 | 10000 | 0.70600 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 16 | 10000 | 0.71400 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 17 | 94740 | 0.67700 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 18 | 95840 | 0.70500 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 19 | 95840 | 0.70700 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 20 | 95840 | 0.70500 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 21 | 88000 | 0.81000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 22 | 82200 | 0.70300 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 23 | 82200 | 0.78700 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 24 | 82200 | 0.70300 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 25 | 77400 | 0.70000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 26 | 81000 | 0.94600 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 27 | 88310 | 0.69500 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |

| NAME | MATL | TOT CAP | TOT HEAT | AVG TEMP | T MELT | H MELT | CURE AT 280 F |
|------|------|------------|------------|------------|--------|--------|---------------|
| SAND | 1 | 1.0739E 00 | 1.0828E 02 | 1.0089E 02 | 0.0 | 0.0 | 0.0 |
| SEL | 2 | 2.8236E 01 | 2.6837E 01 | 8.4930E 01 | 0.0 | 0.0 | 0.0 |
| AIR | 3 | 1.4428E 01 | 1.3313E 01 | 8.4930E 01 | 0.0 | 0.0 | 0.0 |

MATERIAL DATA

DATA DECK 1

TRUMP OUTPUT DATA

```

=====
* MISSILE PROBLEM TND DIMENSIONAL
=====
PRINTOUT CYCLE 80
TOTAL TIME 2.49350E 00
6.11440E 02
AVG TEMP 8.72991E 01
HEAT CAPACITY 1.35739D 00
HEAT FLOW 3.96811E 03
HEAT CONTENT 1.18498E 02
TMD SLOW 0
TMD FAST 11
TMD SLOW 0
TMD FAST 11
KWIT 0
DELTMX 1.00000E 12
SMALL 1.00000E 00
FLUX RATE 6.48977E 00
FLUX RATE 4.78108E 00
TEMP FROM GEN 0.0
GEN RATE 0.0
GEN RATE 0.0
TEMP FROM GEN 0.0
=====

```

| NODE | TEMP | DT | DDT | GE N RATE | H | F | CURE AT 280 F |
|------|---------|----|----------|-----------|------------|------------|---------------|
| 1 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 2 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 3 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 4 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 5 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 6 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 7 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 8 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 9 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 10 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 11 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 12 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 13 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 14 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 15 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 16 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 17 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 18 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 19 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 20 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 21 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 22 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 23 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 24 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 25 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 26 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 27 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 28 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 29 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 30 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 31 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 32 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 33 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |
| 34 | 0.9365D | 02 | -0.2991D | 0.0 | 0.3041E-05 | 0.6608E-06 | 0.0 |

```

=====
MATERIAL DATA
NAME MAT 1
SAND MAT 2
AIR MAT 3
TOT CAP 1.07359E 00
TOT HEAT 9.16649E 01
AVG TEMP 8.86444E 01
TMELT 0.0
HMELT 0.0
=====

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DATA DECK 1

TRUMP OUTPUT DATA

* MISSLE PROBLEM TWO DIMENSIONAL
PRINTOUT CYCLE 100 TIME STEP 3.2434E 00
TOTAL TIME 7.84895E 02
AVG TEMP 5.62265E 01
DELIMX 1.0000E 12
SMALL 1.0000E 00
TVPARY 1.0000E 00
NUTS 3
HEAT FROM FLUX -1.52009E 04
FLUX RATE -2.62882E 01
TEMP RATE -1.93668E 01
HEAT GEN 0.0
TEMP FROM GEN 0.0
GEN RATE 0.0
HEAT CONTENT 1.17042E 02
TOO SLOW 0
HEAT FLOW -2.06335E 04
TOO FAST 11
TIME STEP 3.2434E 00
HEAT CAPACITY 1.35739E 00

| NODE | TEMP | DT | DI | DDT | GEN RATE | H | F | CURE AT 280 F |
|------|---------|---------|---------|---------|----------|---------|---------|---------------|
| 1 | 0.87290 | 0.19000 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 2 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 3 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 4 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 5 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 6 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 7 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 8 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 9 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 10 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 11 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 12 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 13 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 14 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 15 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 16 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 17 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 18 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 19 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 20 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 21 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 22 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 23 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |
| 24 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | |

| NOTE | TEMP | DT | DI | DDT | GEN RATE | H | F | CURE AT 280 F |
|------|---------|---------|---------|---------|----------|---------|---------|---------------|
| 25 | 0.16600 | 0.10000 | 0.00000 | 0.20830 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 26 | 0.87290 | 0.19100 | 0.22000 | 0.38870 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |
| 27 | 0.92650 | 0.43280 | 0.00000 | 0.13330 | 0.00000 | 0.00000 | 0.00000 | 0.00000 |

MATERIAL DATA

| NAME | MATL | TOT CAP | TOT HEAT | AVG TEMP | TMELT | HMELT |
|------|------|--------------|--------------|-------------|-------|-------|
| SAND | 1 | 1.0739E 00 | 9.25179E 01 | 8.62137E 01 | 0.0 | 0.0 |
| STL | 2 | 2.8236E -01 | 2.43549E 01 | 8.62137E 01 | 0.0 | 0.0 |
| AIR | 3 | 1.44246E -03 | 1.30603E -01 | 9.0547E 01 | 0.0 | 0.0 |

DATA DECK 1

TRUMP OUTPUT DATA

***** MISSILE PROBLEM TWO DIMENSIONAL *****
 PRINTOUT 8 CYCLE 120 TOO FAST 11 TOO SLOW 0 KWT 0 DELTHX 1.00000E 12 SMALL 1.00000E 00 TVARY 1.00000E 00 NUTS 3
 TOTAL TIME 8.51641E 02 TIME STEP 4.38834E 00 HEAT FLOW -2.50600E 04 FLUX RATE -2.70771E 01 TEMP RATE -1.99480E 01
 AVG TEMP 8.88432E 01 HEAT CAPACITY 1.35739D 00 HEAT CONTENT 1.20594E 02 GEN RATE 0.0 HEAT GEN 0.0 TEMP FROM GEN 0.0

| NODE | TEMP | DT | DDT | GE N RATE | W | H | F | CURE AT 280 F |
|------|-------|---------|---------|-----------|------------|------------|------------|---------------|
| 1 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 2 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 3 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 4 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 5 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 6 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 7 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 8 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 9 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 10 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 11 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 12 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 13 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 14 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 15 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 16 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 17 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 18 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 19 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 20 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 21 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 22 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 23 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 24 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 25 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 26 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 27 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 28 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 29 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 30 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 31 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 32 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 33 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 34 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 35 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 36 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 37 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |
| 38 | 87200 | 0.88920 | 0.20260 | 0.0 | 0.8332E-05 | 0.7702E-06 | 0.1147E-02 | 0.0 |

***** MATERIAL DATA *****
 NAME MATL 1
 SAND 2
 STEEL 3
 AIR 4
 TOT CAP 1.7359E 03
 TOT HEAT 9.4533E 01
 TOT MELT 2.5919E 01
 TOT MGMT 1.4332E 01
 AVG TEMP 8.80574E 01
 TMELT 0.0
 HMELT 0.0

TRUMP OUTPUT DATA

DATA DECK 1

| MISSILE PROBLEM TWO DIMENSIONAL | | | | | | | | | |
|---------------------------------|---------------|--------------|----------------|--------------|----------------|---------------|---------------|------|-----|
| PRINTOUT | CYCLE | TOO FAST | TOO SLOW | KWIT | DELTMX | SMALL | TVARY | NUTS | |
| 12 | 200 | 11 | 0 | 0 | 1.0000E 12 | 1.33330E 00 | 1.33330E 00 | 4 | |
| TOTAL TIME | TIME STEP | HEAT FLOW | HEAT FROM FLUX | FLUX RATE | TEMP FROM FLUX | TEMP RATE | TEMP FROM GEN | | |
| 1.30776E 03 | 6.30006E 00 | -2.75009E 04 | -2.02682E 04 | -2.02682E 04 | -1.54922E 01 | | | | |
| AVG TEMP | HEAT CAPACITY | HEAT CONTENT | GEN RATE | GEN GEN | TEMP GEN | TEMP FROM GEN | | | |
| 1.11887E 02 | 1.35739D 00 | 1.60018E 02 | 0.0 | 0.0 | 0.0 | 0.0 | | | |
| 1 | 13590 | 0.4770 | 0.0 | 0.0 | 0.3440 | 0.2619 | 0.0 | 0.0 | 0.0 |
| 2 | 10590 | 0.4770 | 0.0 | 0.0 | 0.3440 | 0.2619 | 0.0 | 0.0 | 0.0 |
| 3 | 10590 | 0.4770 | 0.0 | 0.0 | 0.3440 | 0.2619 | 0.0 | 0.0 | 0.0 |
| 4 | 10710 | 0.3640 | 0.0 | 0.0 | 0.3696 | 0.3099 | 0.0 | 0.0 | 0.0 |
| 5 | 10590 | 0.3883 | 0.0 | 0.0 | 0.3696 | 0.3099 | 0.0 | 0.0 | 0.0 |
| 6 | 10400 | 0.2010 | 0.0 | 0.0 | 0.2522 | 0.2242 | 0.0 | 0.0 | 0.0 |
| 7 | 10300 | 0.0910 | 0.0 | 0.0 | 0.1261 | 0.1121 | 0.0 | 0.0 | 0.0 |
| 8 | 10200 | 0.0455 | 0.0 | 0.0 | 0.0630 | 0.0560 | 0.0 | 0.0 | 0.0 |
| 9 | 10100 | 0.0227 | 0.0 | 0.0 | 0.0315 | 0.0280 | 0.0 | 0.0 | 0.0 |
| 10 | 10000 | 0.0113 | 0.0 | 0.0 | 0.0157 | 0.0140 | 0.0 | 0.0 | 0.0 |
| 11 | 9900 | 0.0057 | 0.0 | 0.0 | 0.0078 | 0.0070 | 0.0 | 0.0 | 0.0 |
| 12 | 9800 | 0.0028 | 0.0 | 0.0 | 0.0039 | 0.0035 | 0.0 | 0.0 | 0.0 |
| 13 | 9700 | 0.0014 | 0.0 | 0.0 | 0.0019 | 0.0017 | 0.0 | 0.0 | 0.0 |
| 14 | 9600 | 0.0007 | 0.0 | 0.0 | 0.0009 | 0.0008 | 0.0 | 0.0 | 0.0 |
| 15 | 9500 | 0.0003 | 0.0 | 0.0 | 0.0005 | 0.0004 | 0.0 | 0.0 | 0.0 |
| 16 | 9400 | 0.0002 | 0.0 | 0.0 | 0.0002 | 0.0002 | 0.0 | 0.0 | 0.0 |
| 17 | 9300 | 0.0001 | 0.0 | 0.0 | 0.0001 | 0.0001 | 0.0 | 0.0 | 0.0 |
| 18 | 9200 | 0.0000 | 0.0 | 0.0 | 0.0000 | 0.0000 | 0.0 | 0.0 | 0.0 |
| 19 | 9100 | 0.0000 | 0.0 | 0.0 | 0.0000 | 0.0000 | 0.0 | 0.0 | 0.0 |
| 20 | 9000 | 0.0000 | 0.0 | 0.0 | 0.0000 | 0.0000 | 0.0 | 0.0 | 0.0 |
| 21 | 8900 | 0.0000 | 0.0 | 0.0 | 0.0000 | 0.0000 | 0.0 | 0.0 | 0.0 |
| 22 | 8800 | 0.0000 | 0.0 | 0.0 | 0.0000 | 0.0000 | 0.0 | 0.0 | 0.0 |
| 23 | 8700 | 0.0000 | 0.0 | 0.0 | 0.0000 | 0.0000 | 0.0 | 0.0 | 0.0 |
| 24 | 8600 | 0.0000 | 0.0 | 0.0 | 0.0000 | 0.0000 | 0.0 | 0.0 | 0.0 |
| 25 | 8500 | 0.0000 | 0.0 | 0.0 | 0.0000 | 0.0000 | 0.0 | 0.0 | 0.0 |
| 26 | 8400 | 0.0000 | 0.0 | 0.0 | 0.0000 | 0.0000 | 0.0 | 0.0 | 0.0 |
| 27 | 8300 | 0.0000 | 0.0 | 0.0 | 0.0000 | 0.0000 | 0.0 | 0.0 | 0.0 |

| MATERIAL DATA | | | | | | | | | |
|---------------|-------|-------------|------------|------------|-------|-------|---------------|-----|-----|
| NAME | MATL | TOT CAP | TOT HEAT | AVG TEMP | TMELT | HMELT | CURE AT 280 F | | |
| SAND | 1 | 1.0239E 03 | 1.5478E 01 | 1.0457E 02 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 |
| AIR | 3 | 1.44246E 03 | 1.8300E 01 | 1.0215E 02 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 |
| 1 | 13300 | 0.10000 | 0.16670 | 1.0000E 01 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 |
| 2 | 13750 | 0.25000 | 0.45000 | 1.0000E 01 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 |
| 3 | 10700 | 0.50000 | 0.83340 | 1.0000E 01 | 0.0 | 0.0 | 0.0 | 0.0 | 0.0 |

DATA DECK 1

TRUMP OUTPUT DATA

```

=====
* MISSILE PROGRAM TWO DIMENSIONAL
=====
PRINTOUT CYCLE 220          TIME STEP 2.71381E 00          TVARY 1.00000E 00  NUTS 3
          TOO FAST 11          TOO SLOW 0          DELTMX 1.00000E 12  SMALL 1.00000E 00  TEMP RATE
          1.36145E 03          1.10823E 04          8.16445E 03          5.99687E 00
          HEAT CAPACITY 1.357399 00          HEAT GEN 0.0          TEMP FROM GEN
          1.8861E 02          1.61341E 02          0.0          0.0
=====

```

| NODE | TEMP | DT | DI | DDI | GE N RATE | H | F | CURE AT 280 F |
|------|-------|-----|-----|-----|-----------|---------|---------|---------------|
| 1 | 10920 | 0.0 | 0.0 | 0.0 | 0.0 | 154E-06 | 0.83399 | 0.0 |
| 2 | 10920 | 0.0 | 0.0 | 0.0 | 0.0 | 154E-06 | 0.83399 | 0.0 |
| 3 | 11060 | 0.0 | 0.0 | 0.0 | 0.0 | 277E-01 | 0.27777 | 0.0 |
| 4 | 10990 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 5 | 11330 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 6 | 10910 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 7 | 11270 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 8 | 11220 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 9 | 11920 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 10 | 11280 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 11 | 11920 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 12 | 11220 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 13 | 11230 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 14 | 11350 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 15 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 16 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 17 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 18 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 19 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 20 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 21 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 22 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 23 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 24 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 25 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 26 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 27 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 28 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 29 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 30 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 31 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 32 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 33 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 34 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 35 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 36 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 37 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 38 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 39 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 40 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 41 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 42 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 43 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 44 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 45 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 46 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 47 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |
| 48 | 11260 | 0.0 | 0.0 | 0.0 | 0.0 | 317E-01 | 0.31746 | 0.0 |

| NAME | MAT | TOT CAP | TOT HEAT | AVG TEMP | T MELT | H MELT |
|------|-----|-------------|-------------|-------------|--------|--------|
| SAND | 1 | 1.07359E 00 | 1.26427E 02 | 1.1762E 02 | 0.0 | 0.0 |
| STEL | 2 | 2.82356E 01 | 3.47365E 01 | 1.23024E 02 | 0.0 | 0.0 |
| AIR | 3 | 1.44245E 03 | 1.76975E 01 | 1.22689E 02 | 0.0 | 0.0 |

DATA DECK 1

TRUMP OUTPUT DATA

| * MISSILE PROBLEM | | TWO DIMENSIONAL | | DELTMX | | SMALL | | TVARY | | NUTS | |
|-------------------|-------|-----------------|---------------|-------------|--------------|-----------|----------|---------------|-------------|---------------|-------------|
| PRINTOUT | CYCLE | TIME STEP | HEAT CAPACITY | HEAT FLOW | HEAT CONTENT | GEN RATE | HEAT GEN | TEMP FROM GEN | FLUX RATE | TEMP RATE | TEMP RATE |
| 14 | 240 | 2.72729E 00 | 1.35739D 00 | 1.04046E 04 | 1.59855E 02 | 0.0 | 0.0 | 0.0 | 7.66514E 03 | 5.41337E 00 | 1.00000E 00 |
| TOTAL TIME | | 1.41556E 03 | | 1.17767E 02 | | 0.0 | | 0.0 | | 0.0 | |
| AVG TEMP | | 1.17767E 02 | | 1.59855E 02 | | 0.0 | | 0.0 | | 0.0 | |
| NODE | TEMP | DT | DT | DT | DT | GE N RATE | M | H | F | CURE AT 280 F | |
| 1 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 2 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 3 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 4 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 5 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 6 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 7 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 8 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 9 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 10 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 11 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 12 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 13 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 14 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 15 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 16 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 17 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 18 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 19 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 20 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 21 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 22 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 23 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 24 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 25 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 26 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 27 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 28 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 29 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 30 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 31 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 32 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 33 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 34 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 35 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 36 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 37 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |
| 38 | 11420 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.00000 | 0.0 | |

| MATERIAL DATA | | TOT CAP | | TOT HEAT | | AVG TEMP | | TMELT | | HMELT | |
|---------------|------|-------------|-------------|-------------|-------------|-------------|-------------|-------|-----|-------|-----|
| NAME | MATL | 1 | 2 | 1 | 2 | 1 | 2 | 1 | 2 | 1 | 2 |
| STYL | 2 | 1.07356E 01 | 1.26328E 01 | 1.2659E 02 | 3.34328E 01 | 1.17605E 02 | 1.18407E 02 | 0.0 | 0.0 | 0.0 | 0.0 |
| AIR | 2 | 1.42424E 01 | 1.63385E 01 | 1.63385E 01 | 1.63385E 01 | 1.13269E 02 | 1.13269E 02 | 0.0 | 0.0 | 0.0 | 0.0 |

DATA DECK 1

TRUMP OUTPUT DATA

```

=====
* MISSILE PROBLEM TWO DIMENSIONAL
=====
PRINTOUT CYCLE 249 TOO FAST 11 TOO SLOW 0
          TIME STEP 2.21753E 00
          HEAT CAPACITY 1.3739D 00
          HEAT FROM FLUX 1.04904E 04
          DELTIX 1.05000E 12
          KWI 1.72841E 03
          FLUX RATE 7.28502E 00
          SMALL 1.00000E 00
          IVARY 1.00000E 00
          NUTS 2
          TEMP RATE 5.36695E 00
          HEAT GEN 0.0
          GEN RATE 0.0
          GEN RATE 0.0
          HEAT FROM GEN 0.0
          TEMP FROM GEN 0.0
=====

```

| NODE | TEMP | DT | DDT | GEN RATE | M | H | F | CURE AT 280 F |
|------|-------|----|-----|----------|----|----|----|---------------|
| 1 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 2 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 3 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 4 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 5 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 6 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 7 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 8 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 9 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 10 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 11 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 12 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 13 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 14 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 15 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 16 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 17 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 18 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 19 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 20 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 21 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 22 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 23 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 24 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 25 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 26 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 27 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 28 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 29 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 30 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |

| NODE | TEMP | DT | DDT | GEN RATE | M | H | F | CURE AT 280 F |
|------|-------|----|-----|----------|----|----|----|---------------|
| 1 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 2 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 3 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 4 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 5 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 6 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 7 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 8 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 9 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 10 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 11 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 12 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 13 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 14 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 15 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 16 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 17 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 18 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 19 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 20 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 21 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 22 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 23 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 24 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 25 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 26 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 27 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 28 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 29 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |
| 30 | 11500 | 03 | 02 | 00 | 00 | 00 | 00 | 00 |

```

=====
MATERIAL DATA
=====
NAME MAT 1
SAND MAT 2
STEP MAT 3
STR MAT 3
TOT CAP 03
TOT HEAT 02
AVG TEMP 02
TMELT 0.0
HMELT 0.0
TOT HEAT 01
AVG TEMP 01
TMELT 0.0
HMELT 0.0
=====

```

APPENDIX D

Experimental Data

The data presented in this appendix were obtained from the thermocouples on the rocket motor storage container system located at China Lake, California. The thermocouple output was read out on a Honeywell Electronik 25, 24 channel recorder which had been calibrated at 50, 100 and 150°F. The data was taken on two consecutive, typical summer days (August 1 and 2, 1972) at China Lake. Each thermocouple was read once every 24 minutes. The first set of data presents the storage container temperature at four locations plus three different ways of averaging this data. It also presents the ambient temperature and the approximate time of day. The second set of data presents the surface temperature of the rocket motor and three ways to average this data. It also presents the temperature at the center of the rocket motor and the approximate time of day. Figure 32 shows the location of the thermocouples used to collect this temperature data.

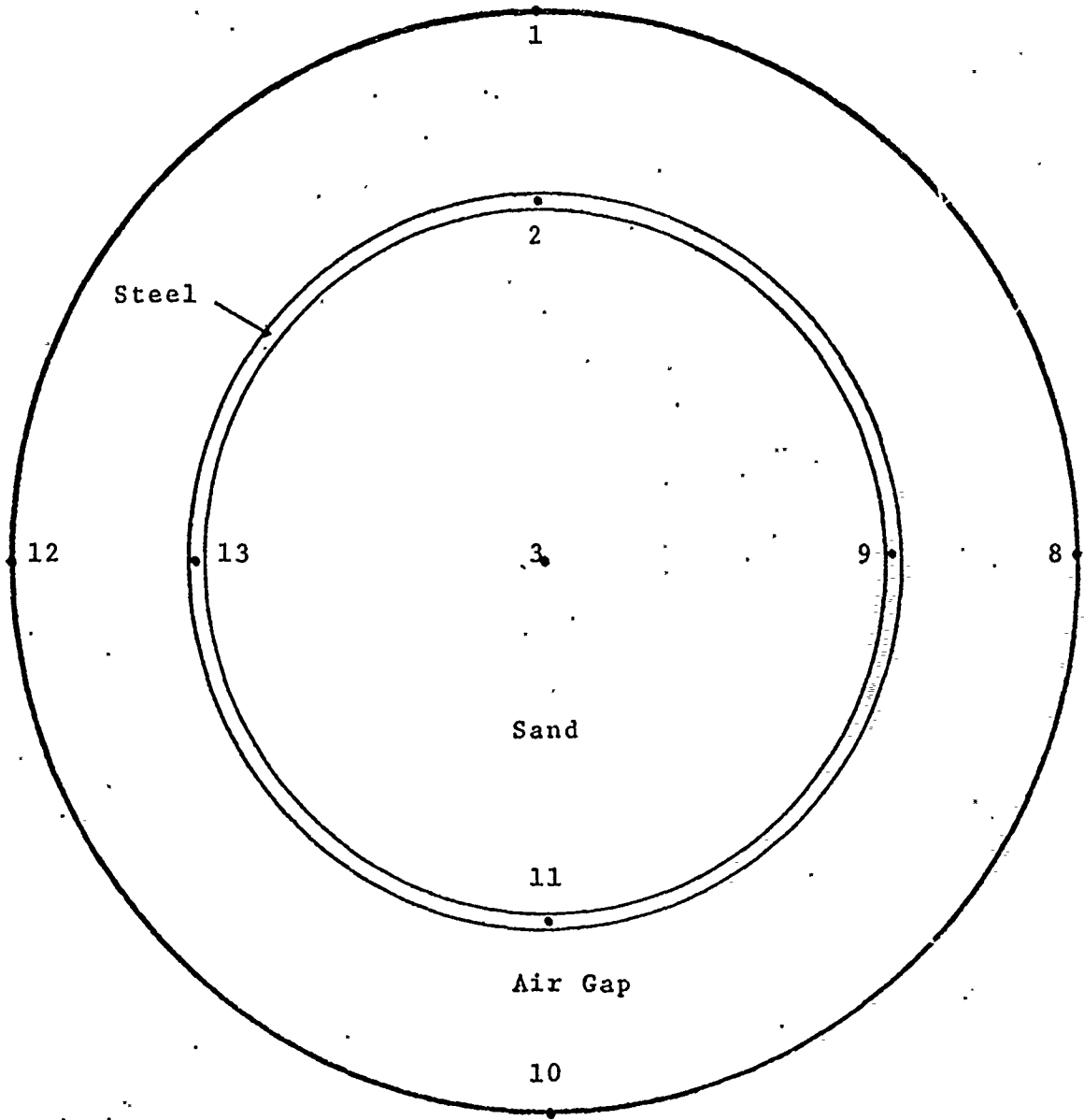


Figure 32: Thermocouple Locations for Experimental Data.

Series 1

| Time (Approximate) Aug. 1, 1972 | Ambient (°F) | #1 (°F) | #8 (°F) | #10 (°F) | #12 (°F) | Avg. #1 & #10 (°F) | Avg. #8 & #12 (°F) | Avg. all 4 "Bulk" |
|------------------------------------|-----------------|------------|------------|-------------|-------------|-----------------------|-----------------------|----------------------|
| 0536 | 76 | 69 | 72 | 79 | 80 | 74 | 76 | 75 |
| 0600 | 77 | 68 | 72 | 79 | 80 | 73.5 | 76 | 74.75 |
| 0624 | 80 | 73 | 75 | 81 | 82 | 77 | 78.5 | 77.75 |
| 0648 | 80 | 81 | 79 | 87 | 85 | 84 | 82 | 83 |
| 0712 | 83 | 90 | 82 | 90 | 89 | 90 | 85.5 | 87.75 |
| 0736 | 85 | 98 | 87 | 93 | 92 | 95.5 | 89.5 | 92.5 |
| 0800 | 87 | 105 | 90 | 94 | 94 | 99.5 | 92 | 95.75 |
| | 89 | 110 | 93 | 96 | 97 | 103 | 95 | 99 |
| | 91 | 116 | 96 | 98 | 99 | 107 | 97.5 | 102.25 |
| | 92 | 121 | 100 | 101 | 102 | 111 | 101 | 106 |
| | 94 | 129 | 103 | 102 | 105 | 115.5 | 104 | 109.75 |
| 1000 | 96 | 133 | 107 | 105 | 108 | 119 | 107.5 | 113.25 |
| | 97 | 139 | 110 | 107 | 110 | 123 | 110 | 116.5 |
| | 100 | 143 | 113 | 107 | 112 | 125 | 112.5 | 118.75 |
| | 101 | 147 | 117 | 109 | 115 | 128 | 116 | 122 |
| | 103 | 150 | 119 | 109 | 116 | 129.5 | 117.5 | 123.5 |
| 1200 | 104 | 153 | 124 | 110 | 119 | 131.5 | 121.5 | 126.5 |
| | 106 | 156 | 128 | 112 | 122 | 134 | 125 | 129.5 |
| | 106 | 157 | 131 | 113 | 124 | 135 | 127.5 | 131.25 |
| | 106 | 154 | 133 | 114 | 125 | 134 | 129 | 131.5 |
| | 107 | 157 | 138 | 115 | 127 | 136 | 132.5 | 134.25 |
| 1400 | 108 | 154 | 143 | 117 | 129 | 135.5 | 136 | 135.75 |
| | 109 | 153 | 142 | 118 | 129 | 135.5 | 135.5 | 135.5 |
| | 110 | 148 | 142 | 119 | 129 | 133.5 | 135.5 | 134.5 |
| | 110 | 143 | 141 | 118 | 128 | 130.5 | 134.5 | 132.5 |
| | 109 | 142 | 143 | 119 | 128 | 130.5 | 135.5 | 133 |
| | 107 | 137 | 143 | 119 | 128 | 128 | 135.5 | 131.75 |
| | 108 | 134 | 139 | 117 | 127 | 125.5 | 133 | 129.25 |
| | 108 | 130 | 139 | 117 | 127 | 123.5 | 133 | 128.25 |
| | 106 | 126 | 138 | 116 | 126 | 121 | 132 | 126.5 |
| | 104 | 123 | 136 | 115 | 125 | 119 | 130.5 | 124.75 |

| Time (Approximate) Aug. 1, 1972 | Ambient (°F) | #1 (°F) | #8 (°F) | #10 (°F) | #12 (°F) | Avg. #1 & #10 (°F) | Avg. #8 & #12 (°F) | Avg. all 4 "Bulk" |
|------------------------------------|-----------------|------------|------------|-------------|-------------|-----------------------|-----------------------|----------------------|
| 1800 | 103 | 118 | 131 | 113 | 123 | 115.5 | 127 | 121.25 |
| | 101 | 113 | 127 | 111 | 120 | 112 | 123.5 | 117.75 |
| | 99 | 107 | 121 | 108 | 117 | 107.5 | 119 | 113.25 |
| | 96 | 101 | 106 | 103 | 111 | 102 | 108.5 | 105.25 |
| 2000 | 95 | 91 | 93 | 98 | 104 | 94.5 | 98.5 | 96.5 |
| | 93 | 88 | 91 | 95 | 101 | 91.5 | 96 | 93.75 |
| | 91 | 86 | 89 | 94 | 99 | 90 | 94 | 92 |
| | 90 | 85 | 88 | 92 | 97 | 88.5 | 92.5 | 90.5 |
| | 89 | 84 | 87 | 91 | 96 | 87.5 | 91.5 | 89.5 |
| | 87 | 83 | 85 | 89 | 94 | 86 | 89.5 | 87.75 |
| 2200 | 87 | 81 | 84 | 88 | 93 | 84.5 | 88.5 | 86.5 |
| | 85 | 80 | 83 | 87 | 91 | 83.5 | 87 | 85.25 |
| | 86 | 79 | 82 | 86 | 91 | 82.5 | 86.5 | 84.5 |
| | 86 | 79 | 82 | 87 | 90 | 83 | 86 | 84.5 |
| | 84 | 79 | 82 | 86 | 89 | 82.5 | 85.5 | 84 |
| 0000 2 Aug. | 82 | 77 | 80 | 85 | 88 | 81 | 84 | 82.5 |
| | 81 | 74 | 78 | 84 | 86 | 79 | 82 | 80.5 |
| | 81 | 72 | 76 | 83 | 85 | 77.5 | 80.5 | 79 |
| | 81 | 72 | 77 | 83 | 85 | 77.5 | 81 | 79.25 |
| | 79 | 72 | 75 | 81 | 83 | 76.5 | 79 | 77.75 |
| 0200 | 79 | 72 | 75 | 80 | 83 | 76 | 79 | 77.5 |
| | 79 | 72 | 74 | 80 | 82 | 76 | 78 | 77 |
| | 78 | 72 | 74 | 79 | 82 | 75.5 | 78 | 76.75 |
| | 77 | 71 | 73 | 79 | 81 | 75 | 77 | 76 |
| | 75 | 70 | 72 | 78 | 80 | 74 | 76 | 75 |
| 0400 | 73 | 67 | 69 | 77 | 78 | 72 | 73.5 | 72.75 |
| | 72 | 66 | 69 | 75 | 77 | 70.5 | 73 | 71.75 |
| | 73 | 65 | 68 | 75 | 76 | 70 | 72 | 71 |
| | 69 | 64 | 67 | 73 | 75 | 68.5 | 71 | 69.75 |
| | 69 | 63 | 66 | 72 | 74 | 67.5 | 70 | 68.75 |
| 0600 | 68 | 63 | 66 | 71 | 73 | 67 | 69.5 | 68.25 |
| | 71 | 65 | 67 | 75 | 76 | 70 | 71.5 | 70.75 |
| | 77 | 74 | 71 | 80 | 79 | 77 | 75 | 76 |
| | 79 | 84 | 77 | 84 | 84 | 84 | 80.5 | 82.25 |

| Time (Approximate) Aug. 2, 1972 | Ambient (°F) | #1 (°F) | #8 (°F) | #10 (°F) | #12 (°F) | Avg. #1 & #10 (°F) | Avg. #8 & #12 (°F) | Avg. all 4 "Bulk" |
|------------------------------------|-----------------|------------|------------|-------------|-------------|-----------------------|-----------------------|----------------------|
| 0800 | 80 | 93 | 81 | 89 | 87 | 91 | 84 | 87.5 |
| | 83 | 101 | 85 | 92 | 91 | 96.5 | 88 | 92.25 |
| | 85 | 108 | 89 | 93 | 93 | 100.5 | 91 | 95.75 |
| | 86 | 114 | 93 | 96 | 97 | 105 | 95 | 100 |
| | 88 | 121 | 96 | 98 | 100 | 109.5 | 98 | 103.75 |
| 1000 | 89 | 126 | 98 | 98 | 101 | 112 | 99.5 | 105.75 |
| | 91 | 130 | 101 | 100 | 103 | 115 | 102 | 108.5 |
| | 94 | 137 | 106 | 102 | 106 | 119.5 | 106 | 112.75 |
| | 96 | 143 | 109 | 104 | 109 | 123.5 | 109 | 116.25 |
| | 100 | 146 | 113 | 106 | 111 | 126 | 112 | 119 |
| | 100 | 149 | 117 | 107 | 114 | 128 | 115.5 | 121.75 |
| 1200 | 101 | 151 | 121 | 108 | 117 | 129.5 | 119 | 124.25 |
| | 103 | 150 | 127 | 110 | 119 | 130 | 123 | 126.5 |
| | 104 | 156 | 129 | 110 | 121 | 133 | 125 | 129 |
| | 104 | 154 | 133 | 111 | 122 | 132.5 | 127.5 | 130 |
| | 105 | 155 | 137 | 112 | 125 | 133.5 | 131 | 132.25 |
| 1400 | 105 | 156 | 141 | 114 | 126 | 135 | 133.5 | 134.25 |
| | 107 | 149 | 143 | 114 | 127 | 131.5 | 135 | 133.25 |
| | 107 | 149 | 146 | 115 | 128 | 132 | 137 | 134.5 |
| | 106 | 151 | 152 | 118 | 131 | 134.5 | 141.5 | 138 |
| | 107 | 144 | 152 | 120 | 132 | 132 | 142 | 137 |
| 1600 | 108 | 139 | 152 | 119 | 131 | 129 | 141.5 | 135.25 |
| | 107 | 136 | 147 | 120 | 130 | 128 | 138.5 | 133.25 |
| | 107 | 136 | 147 | 121 | 130 | 128.5 | 138.5 | 133.5 |
| | 105 | 131 | 147 | 121 | 130 | 126 | 138.5 | 132.25 |
| | 103 | 124 | 144 | 120 | 128 | 122 | 136 | 129 |
| 1800 | 103 | 117 | 134 | 117 | 124 | 117 | 129 | 123 |
| | 100 | 111 | 119 | 113 | 121 | 112 | 120 | 116 |
| | 98 | 105 | 118 | 108 | 117 | 106.5 | 117.5 | 112 |
| | 95 | 99 | 103 | 103 | 109 | 101 | 106 | 103.5 |
| | 93 | 91 | 93 | 97 | 103 | 94 | 98 | 96 |
| 2000 | 91 | 87 | 90 | 94 | 100 | 90.5 | 95 | 92.75 |
| | 90 | 85 | 88 | 93 | 98 | 89 | 93 | 91 |
| | 88 | 83 | 86 | 91 | 96 | 87 | 91 | 89 |

| Time (Approximate) Aug. 2, 1972 | Ambient (°F) | #1 (°F) | #8 (°F) | #10 (°F) | #12 (°F) | Avg. #1 & #10 (°F) | Avg. #8 & #12 (°F) | Avg. all 4 "Bulk" (°F) |
|------------------------------------|-----------------|------------|------------|-------------|-------------|-----------------------|-----------------------|------------------------------|
| 2200 | 87 | 81 | 85 | 90 | 94 | 85.5 | 89.5 | 87.5 |
| | 85 | 80 | 83 | 88 | 93 | 84 | 88 | 86 |
| | 84 | 78 | 81 | 87 | 91 | 82.5 | 86 | 84.25 |
| | 82 | 75 | 78 | 86 | 89 | 80.5 | 83.5 | 82 |
| | 85 | 74 | 79 | 85 | 88 | 79.5 | 83.5 | 81.5 |
| | 83 | 76 | 78 | 85 | 87 | 80.5 | 82.5 | 81.5 |
| | 81 | 74 | 77 | 84 | 86 | 79 | 81.5 | 80.25 |
| 0000 3 Aug. | 78 | 71 | 75 | 83 | 84 | 77 | 79.5 | 78.25 |
| | 81 | 69 | 74 | 81 | 84 | 75 | 79 | 77 |
| | 79 | 70 | 73 | 81 | 83 | 75.5 | 78 | 76.75 |
| | 79 | 72 | 74 | 80 | 83 | 76 | 78.5 | 77.25 |
| | 78 | 71 | 73 | 79 | 82 | 75 | 77.5 | 76.25 |
| 0200 | 77 | 70 | 72 | 78 | 81 | 74 | 76.5 | 75.25 |
| | 73 | 69 | 71 | 77 | 79 | 73 | 75 | 74 |
| | 72 | 65 | 69 | 76 | 78 | 70.5 | 73.5 | 72 |
| | 70 | 64 | 68 | 74 | 77 | 69 | 72.5 | 70.75 |
| | 67 | 63 | 66 | 73 | 75 | 68 | 70.5 | 69.25 |
| 0400 | 67 | 61 | 65 | 71 | 74 | 66 | 69.5 | 67.75 |
| | 66 | 60 | 64 | 70 | 73 | 65 | 68.5 | 66.75 |
| | 65 | 60 | 63 | 70 | 72 | 65 | 67.5 | 66.25 |
| | 65 | 58 | 61 | 69 | 71 | 63.5 | 66 | 64.75 |

| FIRST DAY'S TEMPERATURE RANGES | | | |
|---------------------------------|------|-------|--------|
| HIGH | 110 | 157 | 143 |
| LOW | 69 | 64 | 67 |
| AVG | 89.5 | 110.5 | 105 |
| | | | 136 |
| | | | 129 |
| | | | 75 |
| | | | 68.5 |
| | | | 102.25 |
| | | | 136 |
| | | | 71 |
| | | | 103.5 |
| SECOND DAY'S TEMPERATURE RANGES | | | |
| HIGH | 108 | 156 | 152 |
| LOW | 65 | 58 | 61 |
| AVG | 86.5 | 107 | 106.5 |
| | | | 135 |
| | | | 132 |
| | | | 71 |
| | | | 63.5 |
| | | | 99.25 |
| | | | 142 |
| | | | 66 |
| | | | 104 |
| | | | 138 |
| | | | 64.75 |
| | | | 101.38 |

Series 2

| Time (Approximate) Aug. 1, 1972 | #3 (°F) | #2 (°F) | #9 (°F) | #11 (°F) | #13 (°F) | Avg. #2 & #11 (°F) | Avg. #9 & #13 (°F) | #13 Avg. all 4 (°F) |
|------------------------------------|------------|------------|------------|-------------|-------------|-----------------------|-----------------------|------------------------|
| 0536 | 97 | 83 | 84 | 85 | 85 | 84 | 84.5 | 84.25 |
| 0600 | 96 | 82 | 83 | 85 | 84 | 83.5 | 83.5 | 83.5 |
| 0624 | 95 | 82 | 83 | 84 | 85 | 83 | 84 | 83.5 |
| 0648 | 94 | 85 | 84 | 86 | 89 | 85.5 | 86.5 | 86 |
| 0712 | 94 | 88 | 86 | 88 | 93 | 88 | 89.5 | 88.75 |
| 0736 | 93 | 91 | 88 | 90 | 96 | 90.5 | 92 | 91.25 |
| 0800 | 92 | 95 | 90 | 91 | 99 | 93 | 94.5 | 93.75 |
| | 91 | 98 | 92 | 93 | 102 | 95.5 | 97 | 96.25 |
| | 91 | 101 | 94 | 95 | 104 | 98 | 99 | 98.5 |
| | 91 | 105 | 96 | 97 | 107 | 101 | 101.5 | 101.25 |
| | 91 | 108 | 98 | 98 | 109 | 103 | 103.5 | 103.25 |
| 1000 | 91 | 112 | 101 | 100 | 112 | 106 | 106.5 | 106.25 |
| | 91 | 115 | 103 | 102 | 113 | 108.5 | 108 | 108.25 |
| | 92 | 118 | 105 | 103 | 115 | 110.5 | 110 | 110.25 |
| | 94 | 120 | 108 | 105 | 117 | 112.5 | 112.5 | 112.5 |
| | 94 | 122 | 110 | 106 | 118 | 114 | 114 | 114 |
| 1200 | 95 | 125 | 111 | 108 | 119 | 116.5 | 115 | 115.75 |
| | 96 | 126 | 114 | 109 | 120 | 117.5 | 117 | 117.25 |
| | 98 | 128 | 117 | 111 | 121 | 119.5 | 119 | 119.25 |
| | 100 | 130 | 119 | 112 | 122 | 121 | 120.5 | 120.75 |
| | 101 | 130 | 120 | 113 | 122 | 121.5 | 121 | 121.25 |
| | 102 | 131 | 122 | 114 | 122 | 122.5 | 122 | 122.25 |
| 1400 | 103 | 131 | 123 | 115 | 121 | 123 | 122 | 122.5 |
| | 105 | 131 | 124 | 117 | 121 | 124 | 122.5 | 123.25 |
| | 107 | 129 | 125 | 117 | 121 | 123 | 123 | 123 |
| | 108 | 129 | 125 | 118 | 120 | 123.5 | 122.5 | 123 |
| 1600 | 109 | 127 | 125 | 117 | 119 | 122 | 122 | 122 |
| | 111 | 126 | 125 | 117 | 119 | 121.5 | 122 | 121.75 |
| | 112 | 125 | 125 | 118 | 118 | 121.5 | 121.5 | 121.5 |
| | 113 | 124 | 125 | 118 | 117 | 121 | 121 | 121 |
| | 114 | 123 | 124 | 117 | 116 | 120 | 120 | 120 |

| Time (Approximate) Aug. 1, 1972 | #3 (°F) | #2 (°F) | #9 (°F) | #11 (°F) | #13 (°F) | Avg. #2 & #11 (°F) | Avg. #9 & #13 (°F) | Avg. all 4 (°F) |
|------------------------------------|------------|------------|------------|-------------|-------------|-----------------------|-----------------------|--------------------|
| 1800 | 115 | 121 | 123 | 117 | 115 | 119 | 119 | 119 |
| | 115 | 119 | 121 | 115 | 113 | 117 | 117 | 117 |
| | 116 | 117 | 119 | 114 | 112 | 115.5 | 115.5 | 115.5 |
| | 117 | 115 | 116 | 112 | 111 | 113.5 | 113.5 | 113.5 |
| | 117 | 111 | 111 | 109 | 108 | 110 | 109.5 | 109.75 |
| 2000 | 117 | 107 | 107 | 106 | 106 | 106.5 | 106.5 | 106.5 |
| | 117 | 105 | 105 | 104 | 104 | 104.5 | 104.5 | 104.5 |
| | 116 | 103 | 103 | 103 | 102 | 103 | 102.5 | 102.75 |
| | 116 | 101 | 101 | 101 | 101 | 101 | 101 | 101 |
| | 115 | 100 | 100 | 100 | 100 | 100 | 100 | 100 |
| 2200 | 114 | 98 | 98 | 98 | 98 | 98 | 98 | 98 |
| | 113 | 97 | 97 | 97 | 97 | 97 | 97 | 97 |
| | 112 | 95 | 95 | 96 | 95 | 95.5 | 95 | 95.25 |
| | 111 | 94 | 95 | 95 | 95 | 94.5 | 95 | 94.75 |
| | 110 | 93 | 94 | 94 | 94 | 93.5 | 94 | 93.75 |
| 0000 2 Aug. | 109 | 92 | 93 | 93 | 93 | 92.5 | 93 | 92.75 |
| | 107 | 91 | 91 | 92 | 92 | 91.5 | 91.5 | 91.5 |
| | 106 | 89 | 90 | 91 | 90 | 90 | 90 | 90 |
| | 105 | 88 | 89 | 90 | 89 | 89 | 89 | 89 |
| | 104 | 87 | 88 | 89 | 89 | 88 | 88.5 | 88.25 |
| 0200 | 103 | 87 | 87 | 88 | 88 | 87.5 | 87.5 | 87.5 |
| | 102 | 86 | 86 | 87 | 87 | 86.5 | 86.5 | 86.5 |
| | 101 | 85 | 86 | 87 | 86 | 86 | 86 | 86 |
| | 100 | 84 | 85 | 86 | 85 | 85 | 85 | 85 |
| | 99 | 83 | 84 | 85 | 85 | 84 | 84.5 | 84.25 |
| 0400 | 97 | 82 | 83 | 84 | 83 | 83 | 83 | 83 |
| | 96 | 81 | 82 | 83 | 82 | 82 | 82 | 82 |
| | 95 | 80 | 80 | 82 | 81 | 81 | 80.5 | 80.25 |
| | 94 | 79 | 80 | 81 | 80 | 80 | 80 | 80 |
| 0600 | 93 | 78 | 79 | 80 | 79 | 79 | 79 | 79 |
| | 92 | 77 | 78 | 79 | 78 | 78 | 78 | 78 |
| | 91 | 76 | 77 | 79 | 80 | 77.5 | 78.5 | 78 |
| | 90 | 75 | 78 | 80 | 84 | 79.5 | 81 | 78.5 |
| | 89 | 82 | 80 | 82 | 88 | 82 | 84 | 80.25 |

| Time (Approximate) Aug. 2, 1972 | #3 (°F) | #2 (°F) | #9 (°F) | #11 (°F) | #13 (°F) | Avg. #2 & #11 (°F) | Avg. #9 & #13 (°F) | Avg. all 4 (°F) |
|------------------------------------|------------|------------|------------|-------------|-------------|-----------------------|-----------------------|--------------------|
| | 88 | 87 | 82 | 85 | 92 | 86 | 87 | 86.5 |
| 0800 | 87 | 90 | 85 | 87 | 96 | 88.5 | 90.5 | 89.5 |
| | 86 | 95 | 87 | 89 | 99 | 92 | 93 | 92.5 |
| | 86 | 98 | 90 | 91 | 102 | 94.5 | 96 | 95.25 |
| | 86 | 102 | 92 | 93 | 105 | 97.5 | 98.5 | 98 |
| 1000 | 86 | 106 | 95 | 95 | 107 | 100.5 | 101 | 100.75 |
| | 87 | 108 | 97 | 95 | 108 | 102 | 102.5 | 102.25 |
| | 87 | 111 | 99 | 98 | 110 | 104.5 | 104.5 | 104.5 |
| | 88 | 115 | 102 | 100 | 113 | 107.5 | 107.5 | 107.5 |
| | 89 | 118 | 104 | 101 | 114 | 109.5 | 109 | 109.25 |
| | 90 | 120 | 107 | 103 | 115 | 111.5 | 111 | 111.25 |
| 1200, | 91 | 122 | 109 | 105 | 116 | 113.5 | 112.5 | 113 |
| | 92 | 123 | 111 | 107 | 117 | 115 | 114 | 114.5 |
| | 94 | 125 | 114 | 108 | 117 | 116.5 | 115.5 | 116 |
| | 95 | 127 | 116 | 109 | 118 | 118 | 117 | 117.5 |
| | 97 | 127 | 118 | 110 | 118 | 118.5 | 118 | 118.25 |
| 1400 | 98 | 128 | 120 | 111 | 119 | 119.5 | 119.5 | 119.5 |
| | 100 | 128 | 121 | 112 | 118 | 120 | 119.5 | 119.75 |
| | 102 | 128 | 122 | 113 | 118 | 120.5 | 120 | 120.25 |
| | 103 | 129 | 125 | 115 | 118 | 122 | 121.5 | 121.75 |
| | 105 | 129 | 126 | 116 | 118 | 122.5 | 122 | 122.25 |
| 1600 | 106 | 128 | 127 | 116 | 118 | 122 | 122.5 | 122.25 |
| | 107 | 127 | 127 | 117 | 117 | 122 | 122 | 122 |
| | 109 | 126 | 127 | 118 | 117 | 122 | 122 | 122 |
| | 110 | 126 | 128 | 119 | 117 | 122.5 | 122.5 | 122.5 |
| | 111 | 124 | 127 | 118 | 116 | 121 | 121.5 | 121.25 |
| 1800 | 112 | 122 | 125 | 117 | 115 | 119.5 | 120 | 119.75 |
| | 114 | 120 | 123 | 117 | 113 | 118.5 | 118 | 118.25 |
| | 114 | 117 | 120 | 114 | 112 | 115.5 | 116 | 115.75 |
| | 115 | 114 | 116 | 112 | 110 | 113 | 113 | 113 |
| | 116 | 110 | 110 | 108 | 107 | 109 | 108.5 | 108.75 |
| 2000 | 116 | 107 | 107 | 106 | 105 | 106.5 | 106 | 106.25 |
| | 116 | 104 | 104 | 103 | 103 | 103.5 | 103.5 | 103.5 |
| | 115 | 102 | 102 | 102 | 101 | 102 | 101.5 | 101.75 |

| Time (Approximate) Aug. 2, 1972 | #3 (°F) | #2 (°F) | #9 (°F) | #11 (°F) | #13 (°F) | Avg. #2 & #11 (°F) | Avg. #9 & #13 (°F) | Avg. all 4 (°F) |
|------------------------------------|------------|------------|------------|-------------|-------------|-----------------------|-----------------------|--------------------|
| | 115 | 100 | 100 | 100 | 100 | 100 | 100 | 100 |
| | 114 | 98 | 99 | 99 | 98 | 98.5 | 98.5 | 98.5 |
| 2200 | 113 | 96 | 97 | 97 | 96 | 96.5 | 96.5 | 96.5 |
| | 112 | 95 | 95 | 96 | 95 | 95.5 | 95 | 95.25 |
| | 111 | 93 | 94 | 94 | 93 | 93.5 | 93.5 | 93.5 |
| | 110 | 92 | 93 | 93 | 93 | 92.5 | 93 | 92.75 |
| | 109 | 91 | 91 | 92 | 91 | 91.5 | 91 | 91.25 |
| 0000 3 Aug. | 108 | 89 | 90 | 91 | 90 | 90 | 90 | 90 |
| | 106 | 88 | 89 | 90 | 89 | 89 | 89 | 89 |
| | 105 | 87 | 88 | 89 | 88 | 88 | 88 | 88 |
| | 104 | 86 | 87 | 88 | 87 | 87 | 87 | 87 |
| | 103 | 85 | 86 | 87 | 87 | 86 | 86.5 | 86.25 |
| .0200 | 101 | 85 | 85 | 86 | 86 | 85.5 | 85.5 | 85.5 |
| | 100 | 84 | 84 | 85 | 85 | 84.5 | 84.5 | 84.5 |
| | 99 | 82 | 83 | 84 | 83 | 83 | 83 | 83 |
| | 98 | 81 | 82 | 83 | 82 | 82 | 82 | 82 |
| | 97 | 80 | 80 | 81 | 81 | 80.5 | 80.5 | 80.5 |
| 0400 | 95 | 78 | 79 | 80 | 79 | 79 | 79 | 79 |
| 0424 | 94 | 77 | 78 | 79 | 78 | 78 | 78 | 78 |
| 0448 | 93 | 76 | 77 | 78 | 77 | 77 | 77 | 77 |
| 0512 | 92 | 75 | 76 | 77 | 76 | 76 | 76 | 76 |

FIRST DAY'S TEMPERATURE RANGES

| | | | | | | | | |
|------|-----|-----|-------|------|-----|-----|-------|--------|
| HIGH | 117 | 131 | 125 | 118 | 122 | 124 | 123 | 123.25 |
| LOW | 91 | 79 | 80 | 81 | 80 | 80 | 80 | 80 |
| AVG | 104 | 105 | 102.5 | 99.5 | 101 | 102 | 101.5 | 101.63 |

SECOND DAY'S TEMPERATURE RANGES

| | | | | | | | | |
|------|-----|-----|-----|-----|------|-------|-------|-------|
| HIGH | 116 | 129 | 128 | 119 | 119 | 122.5 | 122.5 | 122.5 |
| LOW | 86 | 75 | 76 | 77 | 76 | 76 | 76 | 76 |
| AVG | 101 | 102 | 102 | 98 | 97.5 | 99.25 | 99.25 | 99.25 |

APPENDIX E

Uncertainty Analysis

An uncertainty analysis was carried out on both the analytical solution and on a one dimensional TRUMP model of the rocket motor storage container system. In both models, the volumetric heat capacity of the sand (ρc), the conductivity of the sand (k), and the emissivity of the surfaces were each varied by ten percent to determine the sensitivity of the system temperature response to each variation. Although other factors may also be varied, it was theorized that these three had the greatest effect on the heat transfer of the system. These factors were also known with the least accuracy; the maximum uncertainty of each was estimated to be plus or minus ten percent (odds 20 to 1).

In the analytical solution, varying the volumetric heat capacity changed parameter a , varying the emissivity changed parameter β , and varying the conductivity changed both parameters a and β . The effects on each parameter from each variation are given in Table V.

TABLE V
Change in Parameters due to Changes in Thermal Properties

| Change in Property | Change in Parameters |
|--------------------------------|------------------------|
| Volumetric Heat Capacity + 10% | $a + .12$ |
| Volumetric Heat Capacity - 10% | $a - .12$ |
| Emissivity + 10% | $\beta + .49$ |
| Emissivity - 10% | $\beta - .37$ |
| Conductivity + 10% | $a - .11, \beta - .22$ |
| Conductivity - 10% | $a + .13, \beta + .35$ |

Each factor was varied holding the other factors constant. The changes in temperature and time delay were computed from the difference between these new values and those previously obtained from the analytical solution. To obtain uncertainty bounds on the analytical curve, the second power equation [Ref. 16] was used, namely

$$\omega_T = \sqrt{\omega_C^2 + \omega_k^2 + \omega_\epsilon^2}$$

where

ω_T = resulting uncertainty in the calculated temperature due to uncertainties in temperature caused by

ω_C = estimated uncertainty in volumetric heat capacity

ω_k = estimated uncertainty in conductivity

ω_ϵ = estimated uncertainty in emissivity

An identical calculation was carried out to calculate the uncertainty in time delay. The results of these calculations are shown in Figures 12 and 13 for the surface and center of the rocket motor respectively. The uncertainty in temperature varied with time with a maximum variation of $\pm 2.75^\circ\text{F}$ at the center of the motor and a maximum variation of $\pm 1.85^\circ\text{F}$ at the surface of the rocket motor. The time delay varied by ± 31 minutes at the center of the motor and ± 11 minutes at the surface. The actual experimental data was also plotted on these Figures for comparison.

The experimental data also had an uncertainty bound. Three primary factors made up this uncertainty bound; the accuracy of the thermocouple wire ($\pm 1.5^\circ\text{F}$), the readability of the recorder ($\pm 1^\circ\text{F}$), and the variation in temperature

caused by inaccuracy in the placement of the thermocouples ($\pm 1^\circ\text{F}$, estimated). The overall uncertainty in the experimental data was also calculated from the second power equation as

$$\omega_T = \sqrt{\omega_{\text{WIRE}}^2 + \omega_{\text{READ}}^2 + \omega_{\text{PLACE}}^2} \approx 2^\circ\text{F}$$

These uncertainty bounds are also shown in Figures 12 and 13.

A procedure, similar to that used to find the uncertainties of the analytical solution, was used to analyze the resulting uncertainty in the TRUMP numerical calculation. The results of these calculations are shown in Figures 14 and 15. The uncertainty in temperature varied with time with a maximum variation of $\pm 2.95^\circ\text{F}$ at the center of the rocket motor and a maximum variation of $\pm 1.95^\circ\text{F}$ at the surface of the motor. The time delay varied from ± 20 minutes at the center of the motor to ± 9 minutes at the surface of the motor.

On the basis of the propagation of uncertainty analysis, it was determined that the solutions were most sensitive, in order of importance, to changes in the volumetric heat capacity, emissivity, and the conductivity.

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