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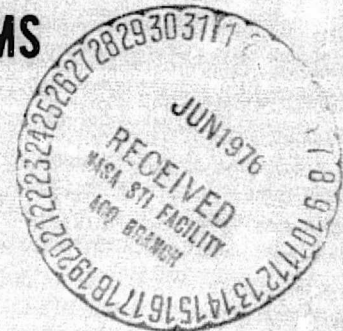
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**ENERGY CONVERSION ALTERNATIVES STUDY**  
**- ECAS -**  
**WESTINGHOUSE PHASE I FINAL REPORT**  
**Volume X — LIQUID-METAL MHD SYSTEMS**

by  
R.R. Holman and T.E. Lippert



**WESTINGHOUSE ELECTRIC CORPORATION RESEARCH LABORATORIES**

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16. Abstract  Plant costs and efficiencies are presented for two basic liquid-metal cycles corresponding to 922 and 1089°K (1200 and 1500°F) for a commercial application using direct coal firing. Sixteen plant designs are considered for which major component equipment were sized and costed. The design basis for each major component is discussed. Also described is the overall systems computer model that was developed to analyze the thermodynamics of the various cycle configurations that were considered.			
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- R. Draper, who developed the design relations for the heat recovery steam generator.
- P. W. Eckels and J. M. Makiel, who provided the design analysis for various heat transfer components.
- R. E. Grumble, who developed and programmed the procedure used to analyze the overall thermodynamic characteristics and for calculating cycle performance. He also contributed significantly to developing the understanding of the physics associated with the liquid metal, two-phase processes.
- J. H. Murphy and F. O. Roach, who developed the required magnet design and model.
- J. L. Steinberg and G. J. Silvestri of the Westinghouse Steam Turbine Division who calculated the performance and price of certain steam turbines.
- G. T. McCreeady and S. M. Scherer of Chas. T. Main, Inc. of Boston, who prepared the balance of plant description and costing, site drawings, and provided consultation on plant island arrangements and plant constructability.

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## SUMMARY

The liquid-metal MHD systems study emphasis is placed on a direct coal fired design using a bubbly two-component flow of sodium and argon in the MHD generator and a Rankine steam bottoming plant. Two basic cycles are studied, corresponding to argon temperatures of 922 and 1089°K (1200 and 1500°F) at the duct inlet. Corresponding to these basic cycles, a total of nine plant designs were sized and costed. The basic systems compresses a two-component bubbly flow of sodium and argon at 8.27 MPa (1200 psi) that expands to about 2.76 MPa (400 psi) in the duct system. The liquid metal and gas are then separated. The argon gives up heat to superheat, reheat and generate steam, and to preheat the combustion air. It is then compressed and passed through the fired heater where its temperature is again increased to 1089°K (1500°F). Lithium-helium is assumed in one plant design.

The MHD duct system that was designed consisted of multiple ducts arranged in clusters and separated by iron magnet pole pieces. The ducts each with an output of about 100 MW were in parallel to the flow but were connected in series electrically to provide a higher MHD voltage. Nonetheless the inversion equipment costs 20% of the total plant cost due to the high currents involved at low-MHD output voltages.

In analyzing the overall systems, the factors that were found to affect most directly the cycle efficiency of this plant are the efficiency of the MHD channel, compressor and pumps, and the pressure losses associated with the large liquid-metal flows. These are areas of great uncertainty, especially in the large component sizes required for commercial operation. The costs determined for both the fired and coupling heat exchanger components were large, comprising nearly 35% of the total capital investment. Due to the large mass of liquid metal circulated, over

63,090 l/s ( $10^6$  gpm) at the 5.51 MPa (800 psi) pressure head, the liquid-metal pump costs and technology are critical system parameters.

With an assumed MHD channel efficiency of 80%, a pump efficiency of 90% and a 45% efficient steam-bottoming plant, the efficiency of the 1089°K (1500°F) combined liquid-metal steam cycle was 43%. The complexity and high cost of this plant (\$1,165/kW) result in a cost of electricity of greater than 12.5 mills/MJ (45 mills/kWh). Lower plant costs can be achieved with lower peak cycle temperatures but overall the energy efficiency is penalized.

With these relatively high plant costs and with the low potential for improving cycle efficiencies over conventional plants, commercial development of this plant would not appear warranted.

## 11. LIQUID-METAL MAGNETOHYDRODYNAMIC SYSTEMS

### 11.1 State of the Art

The liquid-metal magnetohydrodynamic (LM-MHD) power conversion cycle is a developing technology where progress to date has been in the basic theoretical studies of the thermodynamic cycle (s) and small-scale experimental test programs (References 11.1 through 11.4). Probable levels of performance, component technology, and plant costs, however, have not been established for commercial-size systems.

The basic liquid-metal MHD cycle operates by accelerating the liquid through the magnetic field with a gas and, in some instances, maintaining desired velocity by leaving the gas in the liquid for further expansion in the MHD duct and nozzle. Four of the options that exist for embodying this cycle in concept are

- Nozzle fluid acceleration and gas separation (separator type)
- Injector-condenser (two-phase condensing type)
- Bubbly flow (Brayton-type-cycle)
- Slug flow (same principle as bubbly flow).

Present technology and experience have indicated that the two-component, bubbly-flow, Brayton-type cycle (using an inert gas as the drawing fluid and a liquid metal as the prime conversion or electrodynamic fluid) offers the simplest and closest available technology. On this basis, the Brayton cycle was selected for the systems evaluated in this study.

This cycle has been under development in several countries, for several years, but particularly at Argonne National Laboratory, where it was funded primarily by the Office of Naval Research and the National Science Foundation.

The factors that would appear to limit the thermodynamic potential of this cycle are the limits imposed by practical design and safety considerations on liquid-metal and structural temperatures and pressure, and uncertainties associated with the efficiencies of the components.

Operating a liquid-metal system above the atmospheric boiling conditions would impose design safety considerations that lead to significantly increased costs and risk. For sodium and lithium systems a maximum temperature of about 1088°K (1500°F) represents an upper limit that would appear practical. Materials and component design technology to this temperature level are not developed for liquid metals, but the significant advancements being made in the breeder reactor programs should provide a well-founded technology springboard.

The necessity for minimizing liquid-metal vapor carry-over in the gas in the lower temperature portions of the cycle and the relatively high flow void fractions needed within the working fluid to achieve maximum MHD duct efficiencies require high duct inlet pressures, 5.516 to 8.274 MPa (800 to 1200 psi). Such system pressure levels can restrict large-scale designs. This consideration, along with certain low voltage limitations that arise with this particular MHD conversion process, lead to small size, multiple ducts that are operated hydraulically in parallel but are connected electrically in series. This arrangement, however, magnifies the electrical end and friction flow losses for the duct and related components. These practical design considerations suggest that a single, liquid-metal MHD duct be limited to about 100 MWe or less. Large plant sizes (600 to 3000 MWe) would utilize modular duct construction. For such cases the major cost benefit from large-scale plants would be in the economics of the larger heat exchange equipment.

With the emphasis on direct coal-fired systems, the combined LM-MHD steam binary power plant would appear to offer the maximum potential for performance. Within the temperature limits cited above, and based on current steam plant technology, the thermodynamic efficiency of this

combined cycle would range around 45%. Current or very near term emission control technology should be adequate to minimize the environmental impact of this power plant.

The factors that affect the cycle efficiency of this plant most directly are MHD duct efficiency, compressor and pump efficiencies, and the pressure losses associated with the large liquid-metal flows. These are areas of great uncertainty, especially in the large component sizes required for commercial operation, and are subject to varied expert opinions. The principal system technology and component designs, therefore, have been reviewed with regard to the current state of the art which is briefly summarized in Table 11.1. None of the key system components reviewed were found to be sufficiently developed in the required commercial sizes for LM-MHD systems. The results of this study, then, are compared on a relative basis. Specific conclusions on performance and cost are made but should be considered preliminary in that with increased understanding of large-scale LM-MHD component design performance, the various cost tradeoffs would change.

## 11.2 Description of Parametric Points to Be Investigated

### 11.2.1 Selection of Parametric Study Points

Table 11.2 summarizes the LM-MHD cases studied and the major variable(s) that characterize each of the cases. Common assumptions concerning each case are also listed. Eighteen cases are shown, including the base case (Point 16). Thermodynamic state point data and plant efficiencies have been developed for each of these cases. In concurrence with NASA-Lewis technical direction,<sup>\*</sup> however, only ten of these cases have been completely costed. The costed cases are indicated by an asterisk. The noncosted cases represent direct variations of the base case that result in decreased system performance and only a nominal change in the major factors that affect costs. The ten costed cases reflect variation of the parameters that were found to impact most directly on plant performance and cost. These variables include:

---

\* Letter from W. J. Brown (NASA-Lewis) dated April 22, 1975.

Table 11.1 Summary of Status of Liquid-Metal, MHD Key Component Development

Component	Summary of Development Status and Experience
MHD-Duct	<ul style="list-style-type: none"> <li>● Intensive theoretical studies in progress</li> <li>● Most experience with two-component-homogeneous flow</li> <li>● Small-scale tests show duct efficiencies to 50%; Trends in data and theory suggest higher efficiencies are obtainable</li> <li>● Major duct losses are slip and electrical end losses</li> <li>● Materials compatibility at 922°K (1200°F) appears near term, constant temperature operation favorable</li> </ul>
Mixer	<ul style="list-style-type: none"> <li>● No design basis available for high void fractions required</li> </ul>
Nozzle	<ul style="list-style-type: none"> <li>● Test conducted on small nozzles with two-phase flow show efficiencies to 85%</li> <li>● Data on material erosion problem at high temperature required</li> <li>● Application of jet pump at high voids needs to be demonstrated</li> </ul>
Separator	<ul style="list-style-type: none"> <li>● Several design concepts under evaluation</li> <li>● Separator efficiency to 99%, higher recovery may require staged separators</li> <li>● Energy efficiency to 90%</li> <li>● Materials and design untested in commercial-scale systems</li> </ul>
Diffusers	<ul style="list-style-type: none"> <li>● Single-phase flow diffusers design technology established</li> <li>● Achievable efficiency to 85 to 90%</li> </ul>
Liquid-Metal Pumps	<ul style="list-style-type: none"> <li>● Sodium pump technology available at low heads, low flow</li> <li>● Pump costs high; development of direct sodium seals required</li> <li>● Technology extension to 922°K (1200°K) deemed feasible for near term development</li> </ul>
Liquid-Metal Purification	<ul style="list-style-type: none"> <li>● Adequate technology base available</li> <li>● May require larger-scaled system to be economical</li> </ul>
Heat Exchange Equipment	<ul style="list-style-type: none"> <li>● Materials technology to 1088°K (1500°F) needs to be developed for utility applications</li> <li>● Design basis for clean gas systems available</li> <li>● Flue gas/liquid-metal materials compatibility require more cost-effective design and technology base</li> </ul>

Table 11.2 - Summary of Parametric Point Data for LM-MHD Study

Case No.	Combustor	LM-MHD	Bottom Plant
16* Base case (Fig. 11.1) 1000 MWe base load 30 yr life	Direct coal - fired cyclone furnace and stack Illinois No. 6 Coal	Na/A-1200°F, 1200 psi MHD Conversion = 0.75 Duct = 0.75; Pump = 0.85 Compressor Efficiency=0.85 Magnetic Fld. = 0.55T Induction	Steam - 3500 psi 1000°F/1000°F Wet Cooling Tower
1. 600 MWe	Same as base case	Same as base case	Same as base case
2. 3000 MWe	"	"	"
3. 1000 MWe	Atm fluidized bed	"	"
4.* 1000 MWe (Fig. 11.5)	MHD open cycle	"	No steam plant
5.* "	Pres. fluidized bed (100 atm)	"	Same as base case
6.* "	Same as base case	Li/Kc, 1500°F, 1200 psi	"
7. "	"	Na/A, 1200°F, 440 psi	"
8.* " (Fig. 11.4)	"	Na/A, 1500°F, 1200 psi	"
9.* "	"	High-comp. efficiencies	"
10. "	"	Low-comp. efficiencies	"
11.* "	"	Liquid metal electromagnetic pump	"
12.* "	"	Liquid metal jet nozzle pump	"
13.* " (Fig. 11.6)	"	Na/A, 1500°F, 1200 psi	No bottom plant- Once-through cooling
14.* "	"	Na/A, 1500°F, 1200 psi High-component eff.	Same as base case
15. "	"	Same as case 14	Same as base case but higher cycle eff.
16. Base case	See above	See above	See above
17. "	"	Multistaged ducts	"
18.* "	Atm fluidized bed	Same as base case	Same as base case

Assumptions - Common to all Points

Liquid metal aerosol carry-over &lt; 0.1% in gas

Purity level - 2 ppm O<sub>2</sub> and 0.2 ppm H<sub>2</sub>

Gas to liquid-metal reheat in combustor heat exchanger, but no direct LM heating in combustor

Liquid-metal system technology commensurate with liquid-metal fast breeder reactor

MHD output voltage  $\geq 500$  V<sub>dc</sub> for converters to obtain 99% conversion efficiency

System design and component performance with exception of MHD loop will conform to existing central power station practices and all safety requirements

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- MHD duct efficiency
- MHD compressor and pump efficiency
- MHD duct inlet temperature
- MHD working fluids
- Steam bottoming plant efficiency
- Power plant cycle configuration

### 11.2.2 Discussion of Study Parameters

MHD duct performance at the flow rates and component sizes considered for the commercial plants in this study are without direct experimental support and, therefore, require substantial extrapolation. Because of this uncertainty, a range of values (from 70 to 85%) of duct efficiency were assigned for study. The lower numbers (70 to 75%) are values that would appear obtainable in large systems, based on current theory and trends in existing data (References 11.4 through 11.6).

As with any thermodynamic cycle, compressor and pump efficiencies are important performance parameters. Large, highly efficient compressors are state of the art in conventional gas turbine cycles. With the LM-MHD system, however, the effect of liquid-metal carry-over (in the form of an aerosol or oxide) on compressor performance has not been quantified. For these reasons, values of compressor efficiency of 85% and 87.5% are used in the study. Values for the liquid-metal pump efficiency were taken from 80 to 90%. Breeder reactor sodium pump studies indicate efficiencies of around 85% (Reference 11.7). These are single-stage, low-head designs.

The duct inlet temperature and MHD working fluids affect the performance of the cycle in the usual straightforward manner. The impact of these parameters on costs are significant because of component material requirements. Two temperature levels were chosen, 922°K (1200°F) and 1088°K (1500°F). The 922°K (1200°F) is considered as near term technology, but the 1088°K (1500°F) would require a longer range development program. Liquid sodium and argon gas were selected as the working fluids for the base case and all parametric points except for one study case. (For this case, lithium/helium were used). Argon, although requiring significantly larger flow rates, was chosen in consideration of the more limited helium

resource and economics of recovery. Liquid sodium was chosen because of the wider experience and technology base now existing.

### 11.2.3 Description of LM-MHD/Steam Binary Cycle Configuration

The direct coal-fired combined LM-MHD/steam binary power plant was emphasized in the study. Figure 11.1 shows the schematic diagram of this power plant which represents the base case study point.

The major items of the MHD loop are the mixer, the MHD channel (including superconducting magnet), the nozzle-separator, and the liquid-metal primary pumps. The other major loop components include the argon compressor(s) and the interfacing heat exchangers. Not included in the schematic but included as major cost items are the power conversion equipment (inverters, transformers, and circuit breakers) and the liquid-metal auxiliary systems (such as purification, emergency dump, and storage).

The Brayton-type MHD cycle operates by mixing the inert gas and liquid metal at high pressures to form a bubbly, homogeneous flow. The initial void fraction of this mixture is approximately 65%. The mixture is then introduced into the MHD channel, where the gas is allowed to expand through the channel carrying the liquid metal along. The interaction of the flowing liquid metal with the applied magnetic field produces the desired electric field. In the channel the gas phase is expanded to a degree sufficient to produce a two-phase mixture of 85% void.

At the exit of the channel the two components are slightly accelerated, then separated (by momentum differences). Both components then progress separately through the system and are later remixed back at the inlet of the MHD channel.

Prior to being remixed, the liquid metal is pumped back to its MHD inlet head condition, 8.274 MPa (1200 psi). Three methods of liquid-metal pumping were considered in the study:

- Conventional, staged, centrifugal mechanical pumps
- Electromagnetic pumps
- Two-phase nozzle and diffuser.

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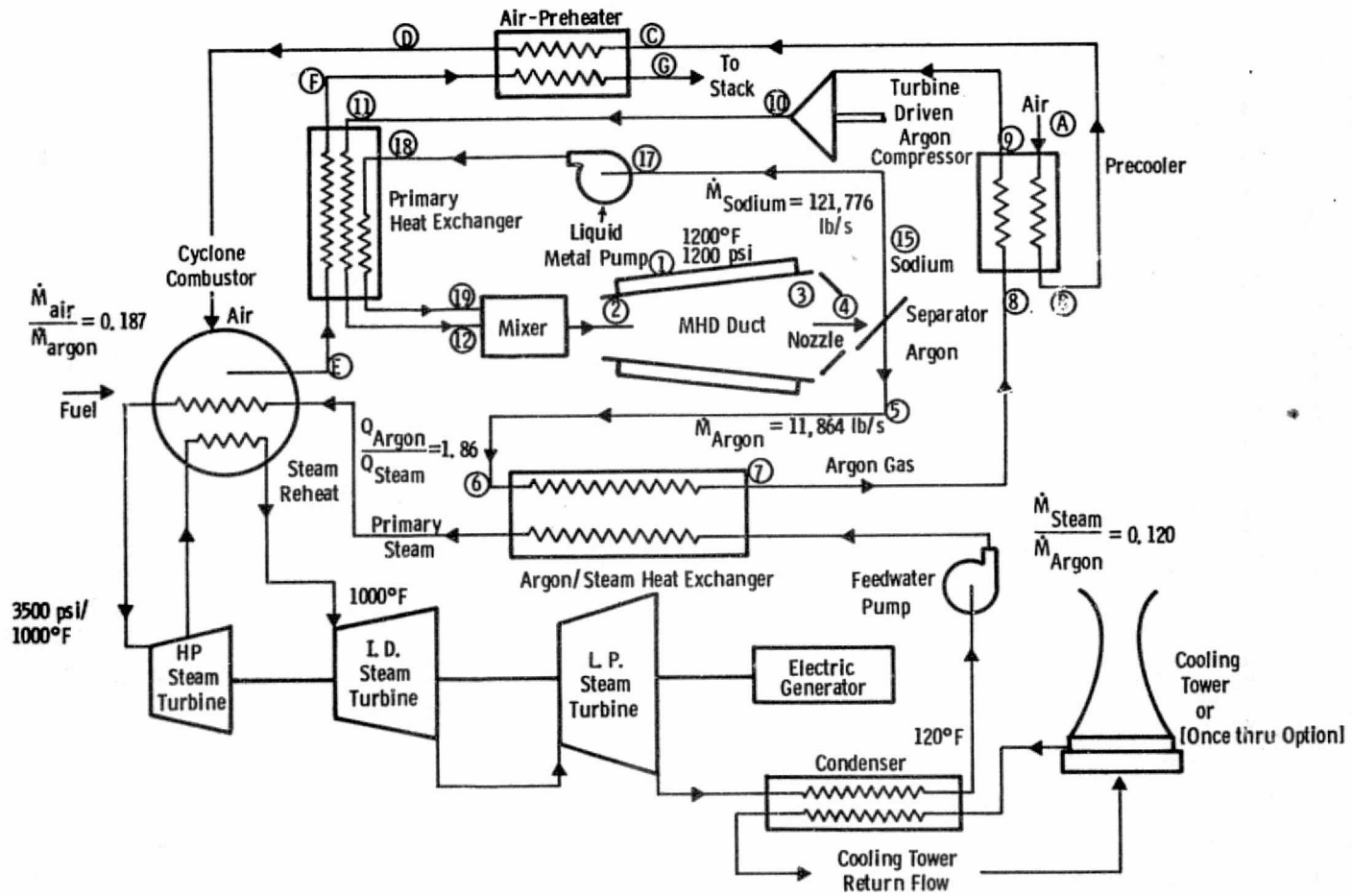


Fig. 11.1—Power plant flow schematic for the base case (Point 16) and Points 1, 2, 6, 7, 9 and 10 (See Appendix A 11.2 for state point values for the base case)

The high-temperature argon gas, after leaving the separator, gives its heat to a conventional 24.131 MPa/811°K/811°K (3500 psi/1000°F/1000°F) steam turbine plant. The gas is then cooled further, compressed, and returned to the heat source (in this case, a conventional, atmospheric cyclone combustor).

For the low-temperature base case [922°K (1200°F) sodium], the steam plant bottoming the MHD cycle not only makes use of the sensible heat of the argon but also receives heat directly from the combustion system. Although this arrangement leads to a more complex thermodynamic cycle, it was found to yield the best plant performance for this low-temperature case. The steam finish superheating [700 to 811°K (800 to 1000°F)] and all the reheating is done directly in the fired part of the system. Approximately 35% of the total steam plant heat load comes directly from the fired system. The steam plant design uses no extracted steam for feed heating but uses argon feedwater heaters instead. This gives a lower steam plant cycle performance, but overall plant performance is higher because of the heat rejection requirements on the LM-MHD cycle side that would occur with extractive feedwater heating.

Figure 11.2 shows the respective heating and cooling curves for the two working fluids (argon and water) at the interface(s) between the two cycles—in other words, at the steam generator and cyclone furnace. The argon cooling curve has been approximated by the straight line (i.e., constant specific heat) from 910 to 339°K (1178 to 150°F) corresponding to a 16.7°K (30°F) steam plant condenser approach at 11.852 kPa (3.5 in Hg) abs.

As seen from the figure, however, this 16.7°K (30°F) temperature difference does not represent the limiting pinch-point consideration. The actual pinch point occurs farther along the cooling curve. Because of the approximate nature of both the argon-cooling curve and the water-heating curve [this curve will depend somewhat on pressure drop - 0.6894 MPa (100 psi) assumed herein] no attempt has been made to design to a given pinch-point; instead, end conditions were chosen that would be nominally

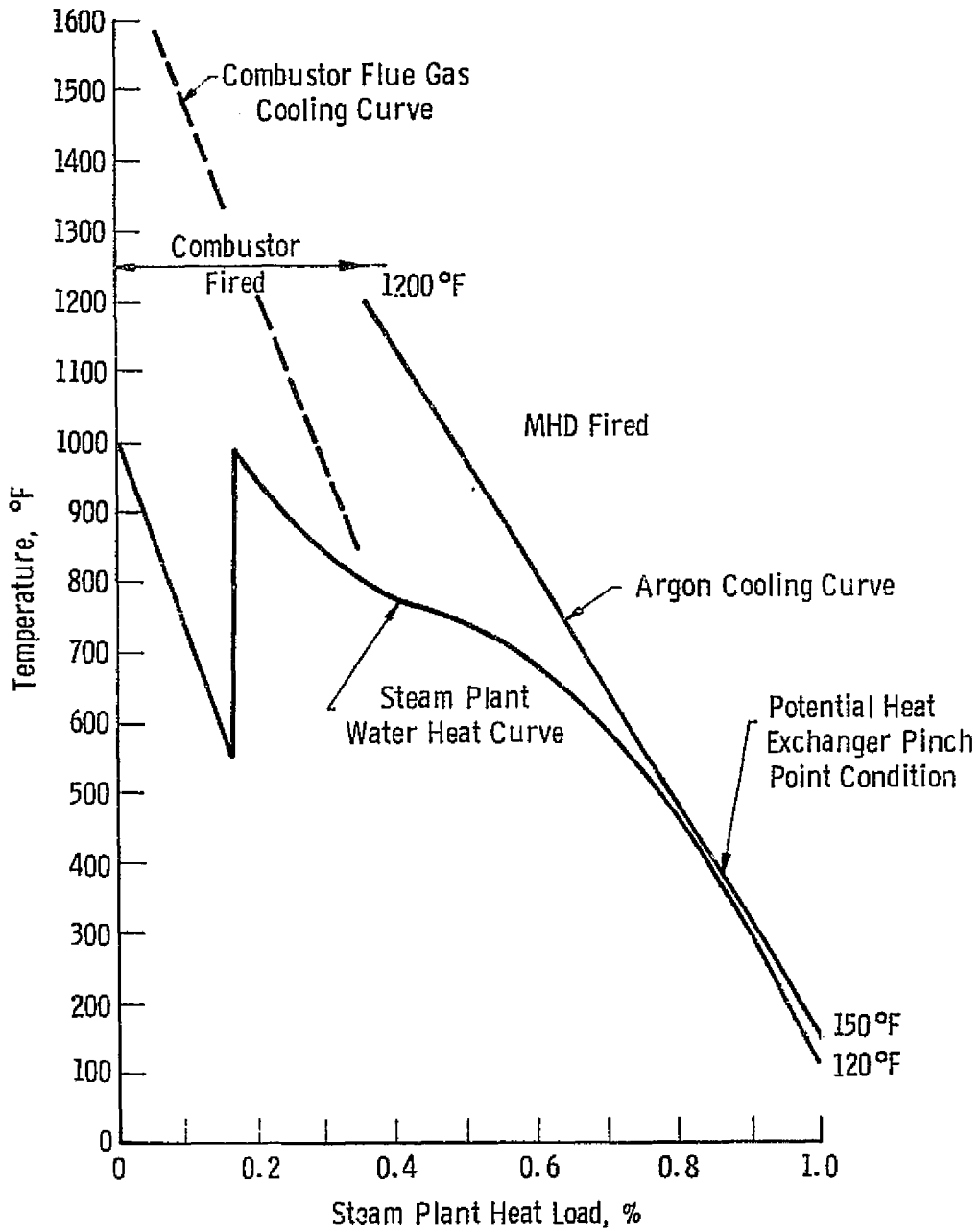


Fig. 11.2—Steam plant heat load curve for LM-MHD topping cycle and supplementary steam plant firing. Base case

acceptable and representative of a reasonable design approach.

For this base case design, single-stage compression preceeded by an argon precooler was found to give the best system performance. The argon precooler serves as the combustion air preheater. Thus, there is no argon heat that is directly rejected to a water heat sink. Most of the argon precooler heat is returned back to the MHD system. The only portion of the heat that is lost is through the combustor stack losses, the specific quantity being dependent on the boiler efficiency of the combustor.

The compressor inlet temperature is determined by the air to argon mass ratio and the design effectiveness of the precooler. The air to argon mass ratio is determined by interfacing conditions at the combustor-primary heat exchanger. These conditions are the desired duct inlet temperature, the approach temperature difference between the argon and flue gas in the primary heat exchanger and the combustion characteristics of the coal. The stack gas temperature will depend primarily on the exit temperature of the flue gas from the primary heat exchanger and the effectiveness of the air economizer.

If there were no direct firing of the steam plant in this base case, the terminal argon temperature would be 450°K (350°F). Various bottom cycle arrangements using this terminal temperature were investigated to determine if higher (than the base case configuration) plant efficiencies could be achieved. Results showed that the heat rejection and a compressor work requirement precluded any efficiency gain.

Figure 11.3 is the schematic diagram for the LM-MHD/steam binary plant for the 1089°K (1500°F) sodium case, and for lithium/helium study point. Because of the higher temperatures, supplementary firing of the steam plant directly from the combustor could be avoided. The terminal argon temperature at the exit of the steam/argon heat exchanger (steam generator) is 380°K (225°F), which leads to a 350°K (170°F) compressor inlet temperature at the precooler exit. The basic elements and components of this cycle are otherwise similar to the low-temperature base case.

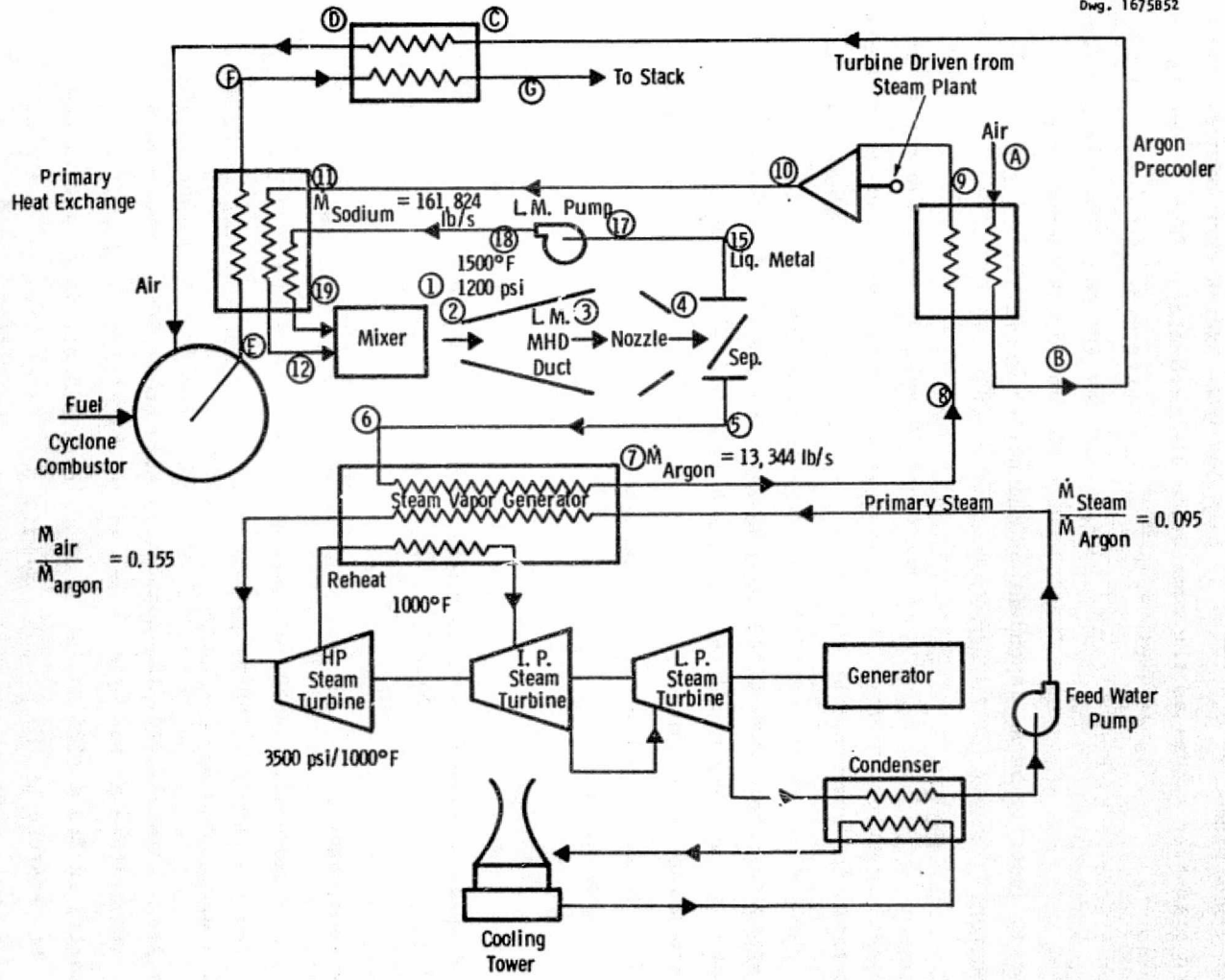


Fig. 11.3—Power plant flow schematic for high temperature parametric cases - Points 8 and 14

11-12

Two of the cases studied in this program were all MHD-type plants. Point 4 uses an open-cycle MHD plant to top the LM-MHD. Figure 11.4 shows a schematic diagram of this all-MHD binary plant. In theory, this combination could lead to relatively high plant efficiencies since, with this combination, the temperature at which energy conversion is occurring has been extended for both the topping and bottoming cycle.

Point 13 is an all LM-MHD plant. Figure 11.5 shows the schematic with this study point; once-through cooling was employed. The components in this plant are essentially those previously identified in the MHD loop of the base case. The major advantage of this all-MHD plant would be its mechanical simplicity. Note in this case that the total flow rate is substantially larger than in the other plant configurations (all are plants based on 1000 MWe net).

In subsequent sections of this report further analysis is made of these power cycles, which includes the determination of plant efficiency and costs.

### 11.3 Approach

#### 11.3.1 Assumptions

The LM-MHD study presented in this section was begun with the presumption of stipulated near state-of-the-art component performance and conversion efficiency (Reference 11.7). MHD duct conversion efficiency and key flow parameters were fixed at estimated attainable, near-optimum values. The study effort was, therefore, originally conceived to place major emphasis on the overall power plant system optimization and performance and cost assessments.

Early in this study, however, the need was established not only to develop some key component concepts for a full-scale system, but also to establish performance estimates in more detail in the MHD loop before any overall system evaluation could be undertaken. The study of interaction of slip ratio, end losses, voltage, aspect ratio, mixer, nozzle



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11-14

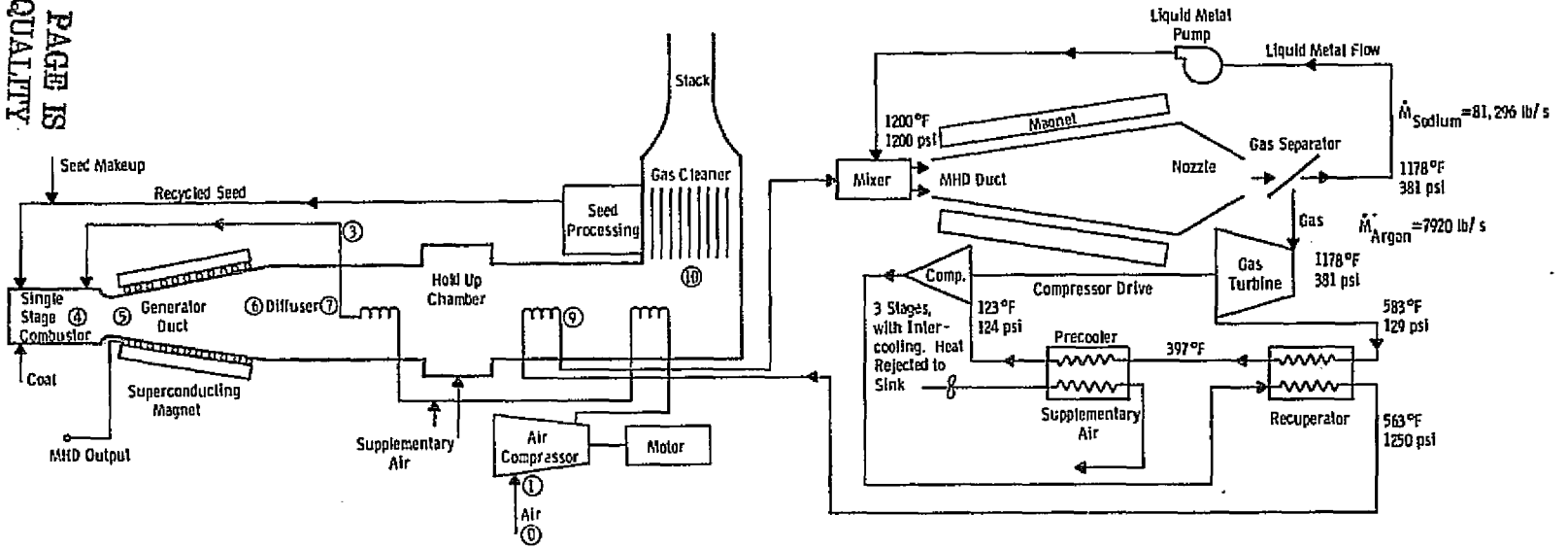
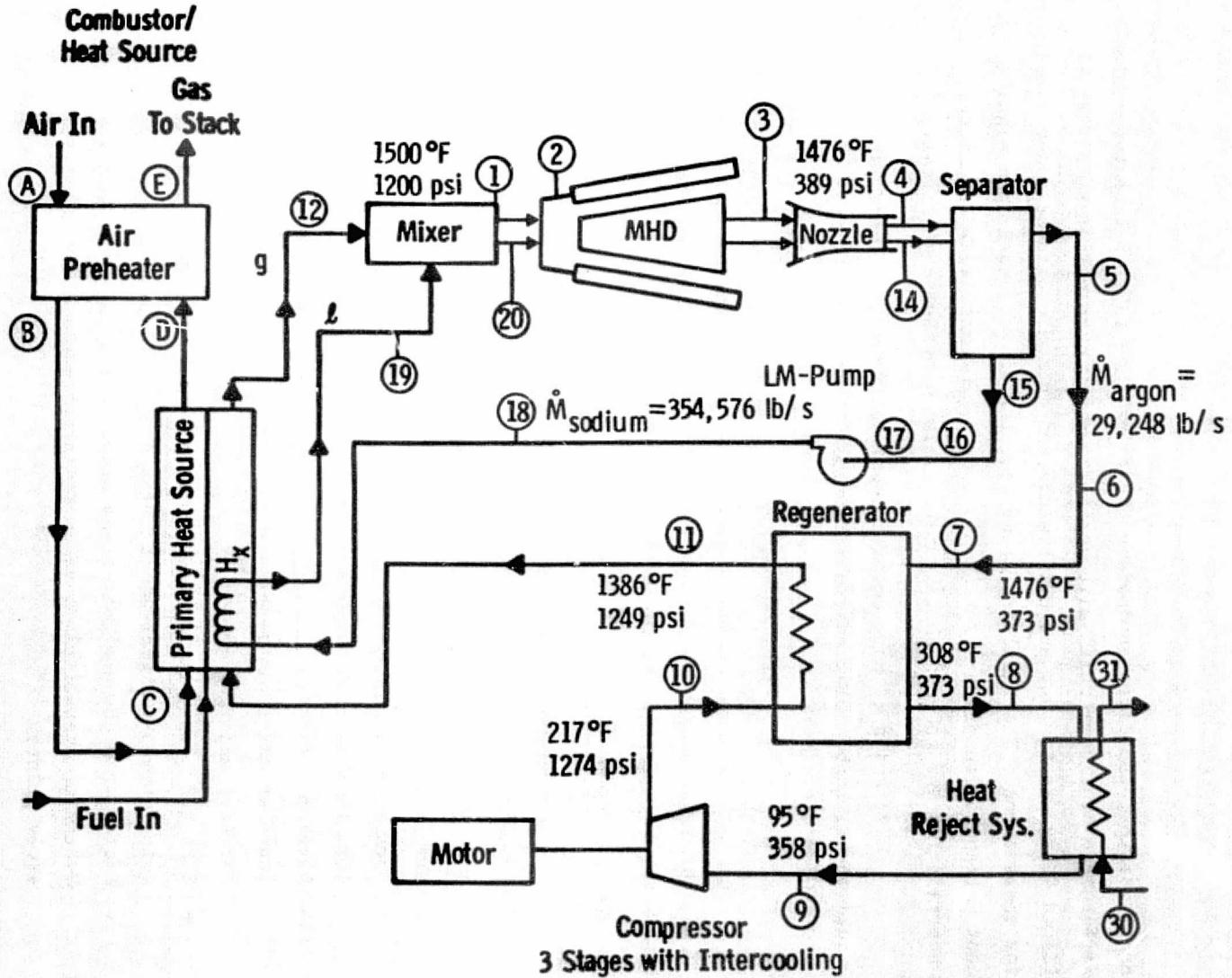


Fig. 11.4—Schematic of open cycle MHD with direct fired air preheater with a two-phase liquid-metal MHD bottom cycle (Point 4)



11-15

Fig. 11.5—Power plant flow schematic for all LM-MHD-Point 13

and separator, and diffuser designs had to be undertaken before the cycle performance of a commercial-size application could be calculated. Established design technology and performance, and economic data were not available for these areas. The resulting study was, therefore, based upon significant extrapolation of data for LM-MHD loop components, with resulting uncertainties and reduced ability to optimize a system realistically.

Discussion of more detailed application of the inert gas, liquid-metal, bubbly flow system and present technology as applied to this study is found in the study results in Appendix A 11.1. Calculational methods, flow models, and property data are considered sufficiently developed for argon and sodium to permit the first task of this study. Subsequent sections and appendices (as identified in the text below) deal with other aspects of the design and evaluation of the system.

#### 11.3.2 Methodology

The methodology used in evaluating the study points in LM-MHD were as follows:

- Evaluate thermodynamic cycle for base case configuration
- Evaluate thermodynamic cycle for the parametric cases
- Conduct preliminary design and costing of major components for the base case study point
- Develop cost data information (capital and installation) based on material, temperatures, and operating conditions, for each major component
- Size and cost the major system components for each parametric point based on algorithms developed from the base case designs
- Factor into the analysis steam plant cost data where applicable
- Determine overall plant costs and performance for the required study points.

The evaluation of the LM-MHD thermodynamic cycle was done with the aid of a computer program. The purpose of the program was to identify the major system components in terms of the relevant thermodynamic and

flow parameters. A description and listing of the program is given in Appendix A 11.2. The major features of the program include

- Incorporation of current analytical model(s) used for describing duct performance
- Cycle configuration flexibility to allow for system optimization, including various options for methods of heat rejection
- Appropriate iterative calculation procedures for interfacing with the bottoming plant to attain desired overall power output.

Included in Appendix A 11.2 is a sample computer output listing (base case computation) showing the calculated parameters.

Because LM-MHD is not a developed technology, the sizing and costing of the major cycle components had to be based on preliminary or conceptual designs. The approach taken was first to develop reasonable design concepts and sufficiently detailed designs for the components that comprise the base case to permit assessment, and then extrapolate these designs to the variations on the base case. This required rather generalized sizing and costing algorithms for each major component.

For instance, the configuration and arrangement of the large-scaled MHD ducts substantially affected the cost not only of the ducts and magnet systems, but of piping and pumping requirements. The initial design efforts performed in the conduct of this study contract have resulted in developing somewhat innovative design concepts that ultimately led to reduced component sizes, piping, and costs. The designs and design basis used for evaluating the base case, Point 16, are summarized in Section 11.5 as is the sizing and costing algorithm used for the MHD duct and assembly and for the major heat transfer equipment. Included are descriptions of the MHD duct assembly that include the mixer, nozzle, and separator; the MHD duct and magnet arrangement; and the preliminary plant layout that identifies the relative arrangement of the major component. The cost basis for determining installation of major components and the materials costing data that were used to establish the capital cost are, also included. Material selection for the various cases was

TABLE 11.3 -LIQUID METAL MHD PARAMETRIC DATA-COSTED CASES

Parametric Point	BC/16	4	6	8	9	11	12	13	14
Power Output, MWe	1900	1000	1000	1000	1000	1000	1000	1000	1000
Fuel - Bituminous (1)	X	X	X	X	X	X	X	X	X
Furnace Type - Fluid Bed Cyclone	C	(3)	C	C	C	C	C	C	C
Conversion Process Parameters									
Gas Flow Rate, lb/s	11646	7922	1251	13200	9918	11341	10301	29254	11349
Liquid Metal Flow Rate, lb/s	119546	81309	85760	160118	101805	116408	105744	354584	137555
Mass Ratio								(6)	
Liquid Metal/Gas	10.26	10.26	68.5	12.12	10.2	10.3	10.3	12.1	12.1
Gas/Steam	8.30	(3)	1.11	10.2	8.30	8.30	8.30	(6)	10.2
Air/Gas	0.189	NA	1.65	0.159	0.199	0.189	0.196	0.082	0.168
MHD Conditions	A/Na	A/Na	He/Li	A/Na	A/Na	A/Na	A/Na	A/Na	A/Na
Duct Inlet Temperature, °F	1200	1200	1500	1500	1200	1200	1200	1500	1500
Duct Inlet Pressure, Atm	82.76	82.76	82.76	82.76	82.76	82.76	82.76	82.76	82.76
Duct Pressure Ratio	3.09	3.09	3.05	3.09	3.09	3.09	3.09	3.09	3.09
Magnetic Flux Density, Telsa	0.55	0.55	1.2	0.45	0.55	0.55	0.55	1.2	0.55
Duct Isentropic Eff.	0.75	0.75	0.75	0.75	0.85	0.75	0.75	0.80	0.80
Compressor Efficiency	0.85	0.85	0.85	0.85	0.875	0.85	0.85	0.85	0.875
Pump Efficiency	0.85	0.85	0.85	0.85	0.90	0.90	0.86	0.85	0.90
Compressor Inlet Temp., °F	120	123	170	175	120	120	120	95 (5)	175
Compressor Pressure Ratio	3.56	4.67	3.51	3.55	3.56	3.56	5.54	1.55 (7)	3.56
Precooler Effectiveness	0.90	0.90 (5)	0.92	0.90	0.82	0.90	0.84	0.92	0.87
Gas ΔP/P (Total-MHD Loop)	0.09	0.08	0.09	0.09	0.09	0.09	0.193	0.111	0.09
Liq. ΔP/P (Total-MHD Loop)	0.06	0.06	0.06	0.06	0.06	0.06	0.03	0.06	0.06
Duct Gross Power, MWe	1229	835 (4)	1549	1645	1155	1197	1087	3864	1498
Cycle Net Power, MWe	45.65	332	148	130.2	187.6	52.0	156	1005	254
MHD Cycle Efficiency	0.0276	0.16	0.063	0.0546	0.122	0.043	0.085	0.364	0.116
Steam Plant/ 3500 psi/ 1000 °F/ 1000 °F									
Gas Temp. Inlet, °F	1178	NA	1488	1488	1178	1178	1178	NA	1476
Steam Temp. Outlet, °F	800	NA	1000	1000	800	800	800	NA	1000
Condenser Press., in Hgabs	3.5	NA	3.5	3.5	3.5	3.5	3.5	NA	3.5
Heat to Steam Plant, MWt	1561	NA	2067	2166	1330	1526	1386	NA	1858
Direct Heat From Combustor, MWt	839.5	NA	0	0	719	825	749.3	NA	0
Net Power, MWe	256	NA	148	0	235	269	0	NA	269
Conversion Cycle Eff.	0.396	NA	0.413	0.401	0.395	0.396	0.396	NA	0.401
Combustor									
Pressure, Atm	1.0	1.0	1.0	1.0	1.0	1.0	1.0	1.0	1.0
Stack Reject Temp., °F	337	305	370	370	337	337	337	370	370
Fuel Rate, tons/hr	465	514	422	438	424	458	435	514	404
Boiler Efficiency	0.88	NA	0.87	0.87	0.88	0.88	0.88	0.87	0.87
Thermodynamic Eff., % (1)	42.0	NA	43.4	42.0	43.9	40.6	42.9	36.4	45.7
Powerplant Eff., %	35.7	30.7	39.1	37.7	39.2	36.3	37.9	32.0	41.0
Overall Eff., %	34.3	29.6	37.5	36.2	37.3	34.5	36.4	31.0	39.4
Total Capital Costs × 10 <sup>-6</sup> , \$	894.5	1318	1728	1118	809.2	854.5	751.1	2039	1106
Capital Costs \$/kWe	942.9	1392	1817	1177	851.7	900.7	790.2	2140	1165
Cost of Elect. Mills/kWh									
Capital	29.81	44.01	57.45	37.21	26.92	28.47	25.0	67.6	36.84
Fuel (2)	8.61	9.88	7.81	8.10	7.83	8.47	8.05	9.51	7.46
Oper. & Maint.	0.94	0.90	0.90	0.92	0.90	0.93	0.91	0.88	0.90
Total	38.35	54.79	56.16	46.23	35.65	37.87	33.94	78.04	45.20
Est. Time of Construction	8.0	9.0	8.0	8.0	8.0	8.0	8.0	8.0	8.0
Est. Availability Date	1983	1983	1990	1990	1983*	1983*	1985	1983	1990

Notes:

\* Assumes Optimistic Performance of Base Case

- ① Where Applicable
- ② Use Base Delivered Fuel Cost
- ③ MHD Open Cycle Topping
- ④ Does Not Include Argon Gas Turbine
- ⑤ Recuperator
- ⑥ All MHD
- ⑦ Per Stage - 3 Stages

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based on the recommendations made in Subsection 3.8. The basis for sizing and costing the superconducting magnet system is given in Appendix A 11.3. The power conditioning equipment description and costs are given in Appendix A 11.4. A description of the required liquid-metal systems and subsystems are given in Appendix A 11.5. Included in this appendix is the basis for determining the appropriate sizing and cost algorithms for:

- Liquid-metal mechanical pump(s)
- EM pump (one design only)
- Liquid-metal purification subsystem
- Liquid-metal inventory and emergency dump.

A summary of the design specifications for the steam generator and primary heat exchanger are given in Appendix A 11.6. (The theoretical basis for the designs is given in Appendix A 9.3).

The factors considered in indirect power plant costs and the subsequent basis for calculating the total plant capitalization and cost of electricity are described in Section 2.

#### 11.4 Results of the Parametric Study

##### 11.4.1 Summary Tables

The schematic diagram previously shown in Figure 11.1 depicts the base case and several low-temperature cases as identified in the title of the figures. The schematic diagrams shown in Figures 11.3 through 11.5 correspond to the cycle configurations for the remaining points. Tables 11.3 and 11.4 summarize the major parameter data, including flows, pressures, temperatures, and efficiencies for all the parametric cases.

The two tables, one corresponding to the required costed points and the second summarizing the noncosted parametric points, have been arranged to show the major variables associated with the topping cycle configuration; major variables associated with the steam bottoming cycle; and the parameters associated with the primary heat source (combustor). The combined-cycle thermodynamic efficiency, power plant efficiency, and

TABLE 11.4 - LIQUID METAL MHD PARAMETRIC DATA-NON COSTED

Parametric Point	1	2	3	5	7	10	15	17
Power Output, MWe	600	3000	1000	1000	600	1000	1000	1000
Fuel - Bituminous (1)	X	X	X	X	X	X	X	X
Furnace Type - Fluid Bed, Cyclone	C	C	FB	FB	C	C	C	C
Conversion Process Parameters								
Gas Flow Rate, lb/s	6962	34812	11576	12081	6834	13914	10898	27600
Liquid Metal Flow Rate, lb/s	71470	392163	118832	124022	191304	142824	132089	120710
Mass Ratio								
Liquid Metal/ Gas	10.26	10.26	10.26	10.26	28	10.26	12.1	13.1 (3)
Gas/ Steam	8.30	8.30	8.30	8.30	8.30	8.30	10.2	10.2
Air/ Gas	0.189	0.189	0.205	0.279	0.189	0.189	0.159	0.323
MHD Conditions								
Duct Inlet Temp., °F	A/Na	A/Na	A/Na	A/Na	A/Na	A/Na	A/Na	A/Na
Duct Inlet Press., Atm	82	82	82	82	30	82	82	82
Duct Press. Ratio	3.08	3.08	3.08	3.08	3.08	3.08	3.08	3.08
Magnetic Flux Density, TESLA	0.55	0.55	0.55	0.55	0.55	0.55	0.55	0.55
Duct Isentropic Eff.	0.75	0.75	0.75	0.75	0.75	0.70	0.80	0.75
Compressor Efficiency	0.85	0.85	0.85	0.85	0.85	0.81	0.85	0.85
Pump Efficiency	0.85	0.85	0.85	0.85	0.85	0.75	0.85	0.85
Compressor Inlet Temp., °F	117	117	115	150	117	120	175	120
Compressor Press. Ratio	3.56	3.56	3.56	3.56	3.52	3.56	3.56	3.56 (3)
Precooler Effectiveness	0.90	0.90	0.90	NA	0.90	0.90	0.90	0.90 (4)
Gas ΔP/ P (Total-MHD Loop)	0.09	0.09	0.09	0.09	0.09	0.09	0.09	0.09
Liq. ΔP/ P (Total-MHD Loop)	0.06	0.06	0.06	0.06	0.06	0.06	0.06	0.06
Duct Gross Power, MWe	734	3670	1220	1274	716	1379	1439	1151 (3)
Cycle Net Power, MWe	29.8	149	50.9	10.1	31.4	-142	188.6	405 (5)
MHD Cycle Efficiency	0.0296	0.0296	0.0310	0.006	0.031	NA	0.092	0.118
Steam Plant/ 3500 psi/ 1000°F/ 1000°F								
Gas Temp. Inlet, °F	1178	1178	1178	1178	1192	1180	1476	1452
Steam Temp. Outlet, °F	1000	1000	1000	1000	1000	1000	1000	1000
Condenser Press., in Hgabs	3.5	3.5	3.5	3.5	3.5	3.5	3.5	3.5
Heat to Steam Plant, MWt	937	4685	1558	1626	931.6	1875	1784	1484
Direct Heat From Combustor, MWt	506	2532	842	879	503.5	1014	0	0
Net Power, MWe	155	774	262	229	167	277	98	0 (6)
Conversion Cycle Eff.	0.396	0.396	0.396	0.396	0.396	0.396	0.45	0.407
Combustor								
Pressure, Atm	1	1	1	100	1	1	1	1
Stack Reject Temp., °F	337	337	327	270	337	337	337	327
Fuel Rate, tons/hr								
Boiler Efficiency	0.88	0.88	0.86		0.88	0.88	0.87	0.87
Thermodynamic Eff., % (1)	40.8	40.8	40.8	39.8	40.2	35.7	48.9	35.0
Power Plant Eff., %	35.4	35.4	35.4	34.6	35.6	32.0	43.9	30.4
Overall Eff., %								
Total Capital Costs × 10 <sup>-6</sup> , \$								
Capital Costs \$/kWh								
Cost of Elec. Mills/kWh								
Capital								
Fuel (2)								
Oper. & Maint.								
Total								
Est. Time of Construction								
Est. Availability Date								

Notes:

- (1) Where Applicable
- (2) Use Base Delivered Fuel Cost
- (3) Per Stage - 3 MHD Stages
- (4) Regenerator
- (5) Total, 3-Ducts
- (6) Steam Plant Power Used for Compressor Drives

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overall efficiency for each case is also shown.

A breakdown of plant costs into total capital and electrical costs (fuel costs, capital, and operating and maintenance) is given for the costed cases (Table 11.3).

#### 11.4.2 Description of Parametric Points

The first column in Table 11.3 corresponds to the base case Point 16. This point represents a technology that was taken to be relatively near term. The MHD cycle is operated at 922°K (1200°F) and 8.27 MPa (1200 psi). The schematic of the cycle was previously shown in Figure 11.1. The power plant bottoming this MHD cycle is a 24.132 MPa/811°F/811°F (3500 psi/1000°F/1000°F) steam turbine plant. This plant, with no extraction feedwater heaters, has a thermodynamic cycle efficiency of about 39.5%.

The power plant efficiency for the combined cycle is about 36%. The cost of electricity for this binary power plant is estimated at 10.91 mills/MJ (39.3 mills/kWh).

The relatively low plant performance for the base case (Point 16) results from selecting relatively modest component efficiencies, assuming a MHD duct efficiency of 75% and assuming 922°K (1200°F) sodium temperature as the near term development capabilities. In this low-temperature case, an austenitic stainless steel was used as the major material for component construction.

Points 9 and 11 show that the base case plant performance increases and plant costs decrease when the efficiencies of the major loop components (compressor and pump) and duct efficiency are increased (see Table 11.3 for specific values). Point 11 uses an EM pump as prime mover for the liquid metal. Point 8 shows that raising duct inlet temperatures to 1088°K (1500°F) increases plant costs significantly. Point 14 shows the combined effect of both higher component and duct efficiencies, as well as a higher (than base case) duct inlet temperature. This plant design showed the highest energy efficiency of the costed study cases.



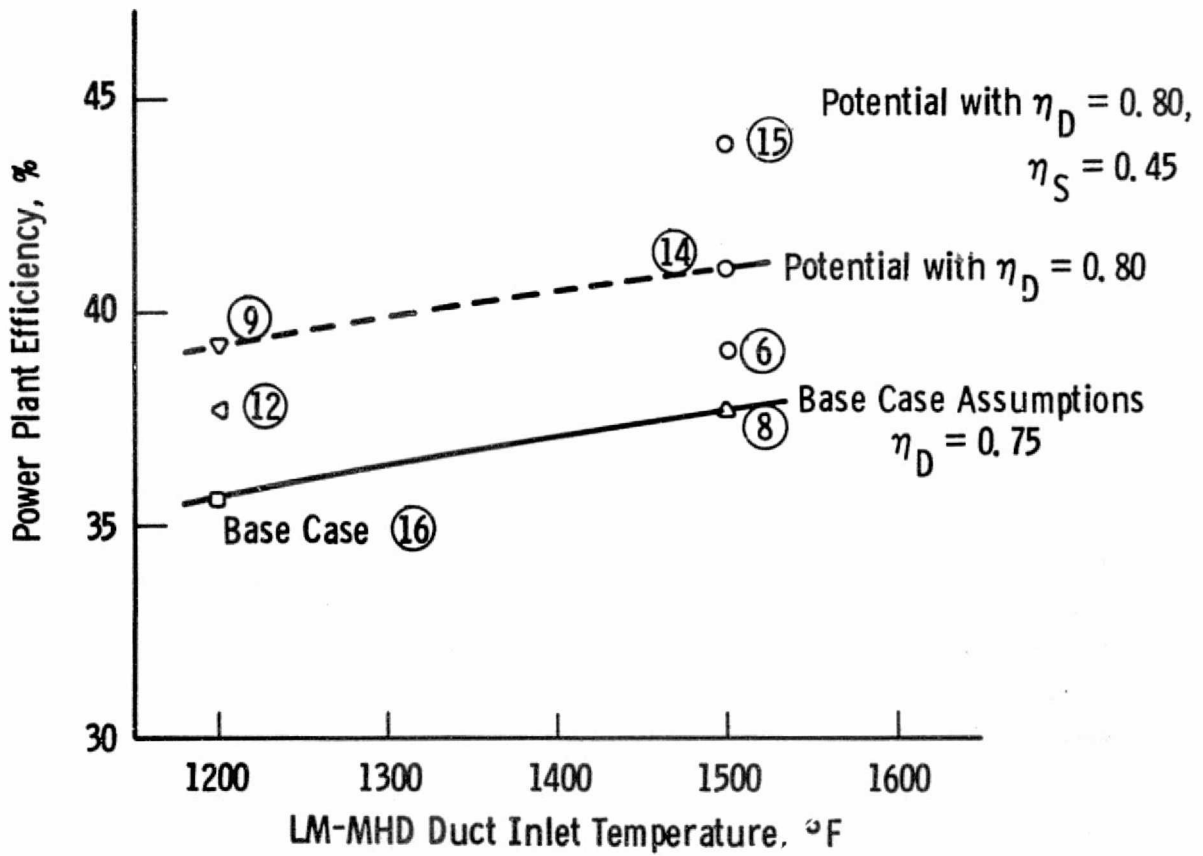


Fig 11.6—Summary and comparison of LM-MHD/ steam binary power plants

Point 12 was a plant design based on replacing the large liquid-metal pumps and drives with a two-phase nozzle diffuser pump. A power plant efficiency increase is obtained in this study case because, like any Brayton cycle, there is a net back work obtained by allowing the gas to work at the higher temperatures (i.e., pump the liquid metal) and by performing work on the gas (i.e., compressing) at the lower cycle temperature. The analysis of this parametric point was made by assuming a two-phase nozzle efficiency of 90% and a corresponding liquid diffuser efficiency of 95%. The combined component isentropic efficiency was then taken to be 86%.

Point 6 uses lithium/helium as the MHD working fluids. The relatively high cost of this plant is the result of constructing many of the loop components from a refractory metal.

Points 4 and 13 represent plant configurations that do not include a steam power plant. Point 4 is a binary plant using an open-cycle MHD power plant to top a liquid-metal MHD plant (see Figure 11.4). The efficiency of this combined cycle was about 31%. This low performance is the result of the relatively poor performance of the liquid-metal MHD bottom plant. Improved performance would be obtained by taking a higher temperature liquid-metal MHD bottom plant and better optimizing the interfacing between the two cycles. Study of this combined all-MHD plant configuration, however, was limited to only one case.

Point 13 is an all LM-MHD power plant. The system shows only marginal plant performance and relatively high plant costs. The high plant costs are due to both the larger duct size and flow requirements and the higher power conversion costs. This result is not unexpected, since the net power output of this system is small compared to the total output of the MHD generator, and parasitic losses have not been reduced.

#### 11.4.3 Analysis of LM-MHD/Steam Binary Plant Performance

Figure 11.6 summarizes the results of the LM-MHD steam binary power plant cases and shows the potential for the system. The figure shows the combined effects of temperature and duct efficiency on power

plant performance. The numbers in the figure correspond to the parametric points listed in Table 11.3. The solid line shown at the bottom represents plant performance over the range of temperatures indicated. This line shows the system potential based on relatively modest assumptions concerning MHD component efficiencies and a duct efficiency,  $\eta_D$ , of 75%.

The dashed line shows the potential performance of the combined cycles, assuming the more optimistic LM-MHD component efficiencies. A major effect here is the assumption of an MHD duct efficiency of 80%.

Point 15 is a noncosted study point where credit is taken for increasing the steam bottoming plant thermodynamic efficiency to 45%. Power plant efficiency of this case is 43%, which represents the highest obtained in the study. The plant costs for this case are expected to correspond approximately to those of Point 14, or about 12.5 mills/MJ (45 mills/kWh).

## 11.5 Capital and Installation Costs of Plant Components

### 11.5.1 Cost Methodology

Nine power plants were evaluated in the LM-MHD study for estimated component capital and installation costs, indirect costs, and balance-of-plant costs. The general approach taken in the evaluation of the indirect and balanced plant cost is discussed in Section 2. The following paragraphs treat the capital and installation costs of the major equipment.

Most of the major components associated with the LM-MHD cycle are not state-of-the-art designs. Preliminary designs of such equipment, therefore, had to be developed before reliable cost estimates could be made. The general methodology that was used to generate the component cost data was as follows:

- Conceptual design of the major equipment assemblies. This involved establishing the MHD duct and magnet arrangement, determining the number of flow loops, identifying the assembly

interfaces, and specifying the requirements for system (or component) redundancy

- Specifications of the design requirements for each major component, such as operating temperature, pressure, flow, stress level, etc.
- Preliminary design of components to meet base case operating requirements
- Derivation of size, weight, and cost algorithms for each major component
- Development of cost data information (capital and installation) based on material selection, temperatures, and required operating conditions.

A summary is given in Table 11.5 of the algorithms that were derived to determine size, weight, and cost of each major component. The nomenclature used in Table 11.5 is defined in Table 11.6. Additional supportive information on these relationships are given in the appropriate appendices. A brief summary of the design and costing basis used for sizing and costing these components is given below.

#### 11.5.2 MHD Duct and Magnet Costs

The MHD duct and magnet components constitute one major assembly requiring design effort to assure realism in both the structural integrity and the cost of the system. The basic assumptions from which the design evolved were:

- Conformity to limitation imposed by current theory on the electrodynamics of the duct
- Use of current liquid-metal technology, where applicable
- Minimization of magnet and duct structure costs
- Maximization of operating life and serviceability.

These considerations ultimately led to the toroidal duct assembly arrangement shown in Figure 11.7. The ducts are oriented vertically and are grouped (three or more) in a circular configuration and interconnected

TABLE 11.5 - SUMMARY OF ALGORITHMS USED IN SIZING AND COSTING POWER PLANT COMPONENTS

Component	Size Algorithms	Weight Algorithms	Cost Algorithms
1. Superconducting Magnet	$V = (0.0997) (D)(L) (k,d)$ $k,d$ = see below		$C_T = (28,573) (V) (V)$
2. Dewar	$A_1 = L_{i,d} \times k_{i,d}$	$W_T = 284 N k (B/S_d) L^2 A_1$	$C_T = W_T C_6$
3. Iron Poles	$k_{i,d}$ = see below	$W_T = \frac{\ln(1+N)}{2} \left( \frac{L}{2n} + \pi \right) L$	$C_T = W_T \times C_5$
4. Structural MID Components		$K = 155 - 17.5 \cdot d$	
Duct	$k_{i,d} = \frac{k_{o,d} + k_{i,d}}{2}$ $A_1 = L \times d$ $A_2 = k_{i,d} \times L$	$W_1 = K (\bar{V}/S_d) k_{i,d}^2 A_1$ $W_2 = K (\bar{V}/S_d) k_{i,d}^2 A_2$	$C_T = 2N (W_1 + W_2) C_1$
Mixer	$A_2 = k_{i,d}^2 \cdot L = k_{i,d}$	$W_T = K (\bar{V}/S_d) k_{i,d}^2 A_2$	$C_T = 4N W_T C_2$
Nozzle	$A_1 = (L)(d)$ $A_2 = (L)(d)$ $L = \left[ 1 - \left( \frac{r_1}{r_0} \right) \left( \frac{V_1}{V_0} \right) \right] \frac{k_{o,d}}{2 \tan \theta_n}$ $k = \frac{k_{o,d} + k_{o,n}}{2}$ $k_{o,n} = \left( \frac{r_1}{r_0} \right) \left( \frac{V_1}{V_0} \right) k_{o,d}$	$W_1 = K (\bar{V}/S_d) k^2 A_1$ $W_2 = K (\bar{V}/S_d) k^2 A_2$	$C_T = 2N (W_1 + W_2) C_3$
Separator/ Purifier	$A_1 = k_{o,n}^2 / \tan \theta_s$ $A_2 = (k_{o,d}) (k_{o,n}) / \tan \theta_s$	$W_1 = K (\bar{V}/S_d) k_{o,n}^2 A_1$ $W_2 = K (\bar{V}/S_d) k_{o,n}^2 A_2$	$C_T = 2N (W_1 + W_2) C_4$
Liquid Metal Return Leg Piping	$L = \sqrt{\frac{m}{\rho} V_p}$ $L_T = L_m + L_G + L_N + L_S$	$W_T = K (\bar{V}/S_d) L^3 L_T N$	$C_T = W_T \times C_7$
5. Liquid Metal Pump	$n = \frac{448.8 \text{ mN}}{2.5 \times 10^3 \cdot \rho}$	$W_T = \frac{(295.2) (m) H}{\rho}$	$C_T = 8 \times 10^6 \cdot n$
6. Liquid Metal Purification	$t = 448.8 \times t \left( \frac{1-p}{p+0.01} \right)$		$C_T = 8572 N (t)^{0.6}$
7. Liquid Metal Inventory & Dump		$W = 1.2 m L_T N / V_p$	$C_T = C_8 W \left( \frac{1063}{\rho L_M} \right) m^{0.7}$
8. Steam Generator	$A_5 = \frac{Q}{HTD} \left[ \frac{d_o}{d_i} \frac{0.4}{0.6} \frac{\beta_1}{\beta_2} \right] \frac{1}{n_2}$ $L = \frac{Q}{HTD} \left( \frac{V d_o}{3600 N m} \right) \frac{1}{\beta_2} \left( \frac{\beta_1}{\beta_2} - 1 \right) \beta_2$ $d = \sqrt{\frac{3600 N m p}{N_1 V (p - \beta_2) \beta_5}}$ $\left. \begin{matrix} \beta \\ V \\ HTD \end{matrix} \right\} \text{ see appendix A 11.6}$	$W_1 = \frac{n d^2 \rho_m}{2} (\bar{V}/S_d) \left( \frac{L d}{2} \right)$ $W_2 = 0.0106 \times d_{o,n} \times C_9 \times A_5$ $W_T = W_1 + W_2$	$C_T = (W_1 C_{10} + W_2 C_{11}) H_1$
9. Primary Heat Exchanger	$d = \sqrt{\frac{28}{\pi} A_c}$ $L = \frac{A_s}{A_c} \left( \frac{d_1}{48} \right)$ $\left. \begin{matrix} A_s \\ A_c \end{matrix} \right\} \text{ see appendix A 11.6}$	$W_T = \frac{6 L_1}{N_1}$	$C_T = N_1 W_T C_{11}$
10. Recuperators Intercoolers Economizers Precoolers	$d = \sqrt{\frac{8.0 A_c}{\pi}}$ $L = \frac{A_c}{A_s} (1.011)$ $\left. \begin{matrix} A_c \\ A_s \end{matrix} \right\} \text{ see appendix A 11.6}$	$W_T = 0.64 T_1$	$C_T = N_1 (L_1 C_9 + N_1 C_{10})$
11. Gas Compressors		$W_T = 144.2 m n_1 + 70,000 \left( \frac{1300}{p} \right)^{0.5}$	$C_T = 10.5 W_T$
12. Gas Piping		$W_T = \frac{2 m \bar{V} \rho_m}{C_9 V} \left( \frac{L_1}{S_d} + \frac{L_2}{S_c} \right)$ $L_1 = \text{Hot Leg Length}$ $L_2 = \text{Cold Leg Length}$	$C_T = W_T C_{12}$

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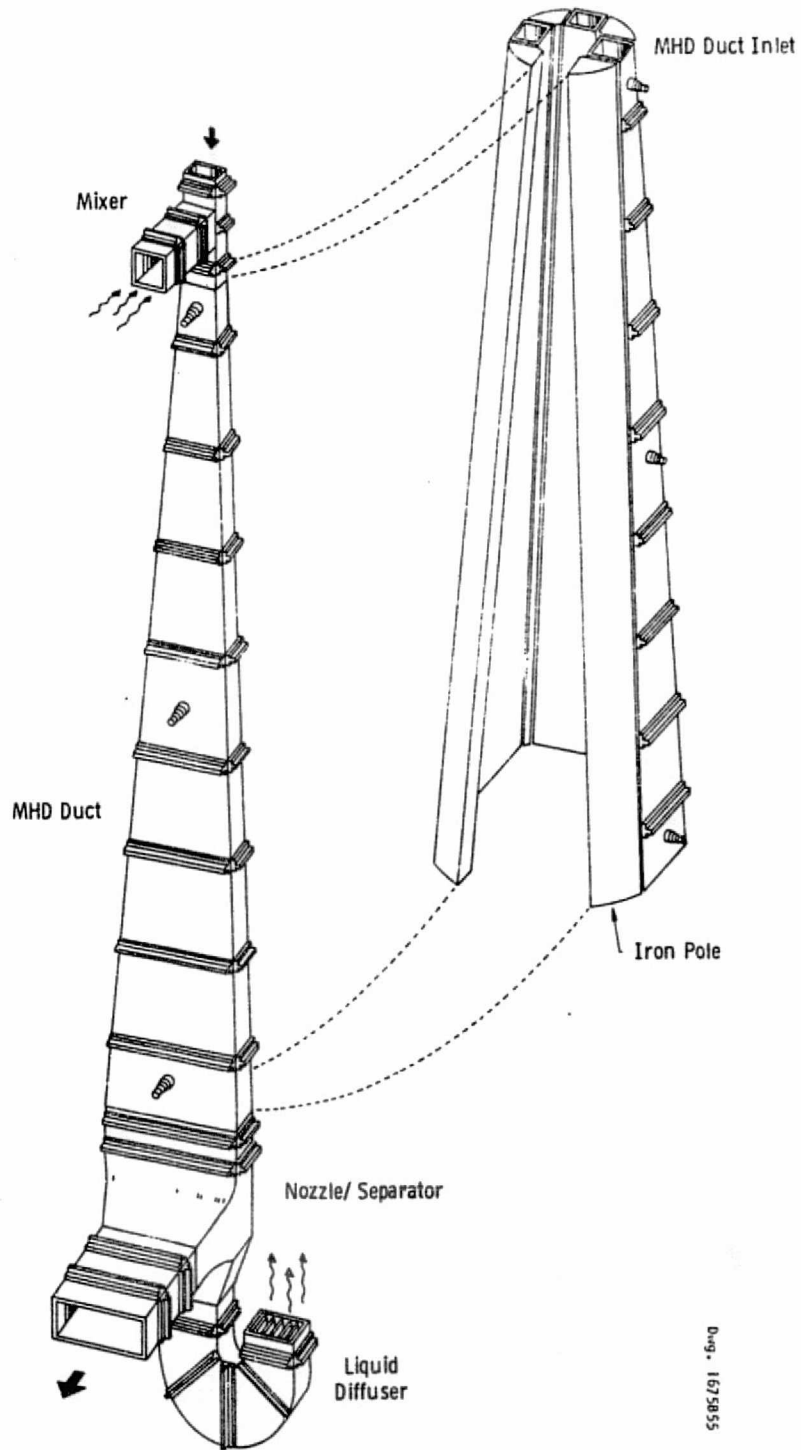
Table 11.6—Definition of Symbols

A	= Surface area, ft <sup>2</sup>
A <sub>C</sub>	= Cross-sectional flow area, ft <sup>2</sup>
A <sub>S</sub>	= Heat transfer surface area, ft <sup>2</sup>
B	= Magnet
C	= Cost Coefficient, \$ / lb or \$ / ft, \$ / tube
C <sub>T</sub>	= Total capital equipment costs, \$
d	= Diameter of heat exchange vessel, ft
d <sub>i</sub>	= Inside tube diameter, in
d <sub>o</sub>	= Outside tube diameter, in
F	= Liquid metal flow, ft <sup>3</sup> /s
f	= Required liquid metal flow rate, gpm
L	= Length of component, ft
ℓ	= Average width or unsupported dimension, ft
L <sub>t</sub>	= Total tubing length, ft
m	= Mass flow rate, lb/s
MTD	= Mean temperature difference, °F
N	= Number of ducts
n	= Number of assemblies or units
N <sub>1</sub>	= Number of heat exchanger units
N <sub>t</sub>	= Number of tubes
p	= Required liquid metal purity, ppm, or tube pitch
$\bar{p}$	= Average pressure, psi
Q	= Total heat load, Btu/hr
S <sub>d</sub>	= Design stress, psi
V	= Fluid velocity, ft/s
W	= Weight, lb
ζ	= Density of Material or Fluid, lb/ft <sup>3</sup>
θ	= Angle

## Subscripts

1	- Varying width plates	m	- Metal or mixer
2	- Constant width plates	n	- Nozzle
d	- Duct	o	- Out
g	- Gas	p	- Pipe
i	- In	s	- Separator
Lm	- Liquid metal		

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Fig. 11.7—LM-MHD duct assembly schematic

with iron pole pieces. Figure 11.8 shows further detail of the design of the ducts, as well as the design and positioning of the magnet with respect to the ducts. The superconducting magnet and dewar sets are fixed to the duct walls adjacent to the iron pole pieces. Thus, the iron pole pieces shape the magnetic field to conform to the geometry. This configuration easily meets the requirements for field uniformity and reduces stray field effects. Further detail concerning design and specification of the magnet and field requirements and the magnet and dewar designs are given in Appendix A 11.3.

The costs of the magnet system were broken into cost of the superconductor including dewar and refrigerators and cost of the iron pole pieces. Conductor costs (a low field, Nb-Ti composition) were taken as \$110/kg (\$50/lb). The cost of the iron pole pieces was based on the total weight of iron used in the design. Iron costs were taken as \$1.58/kg (\$0.72/lb) installed, (Reference 11.8). The algorithm expressing these costs was developed in terms of the MHD duct design parameters. The formulas are given in Table 11.5.

The actual MHD duct portion of the assembly was costed on the basis of the design that was previously shown in Figure 11.8. Designs and costs for the mixer, nozzle, separator-diffuser, and liquid-metal return-leg piping were based on the ribbed structural configuration used for the duct design. The major items considered in the costing of the MHD duct were the materials used for structural pressure housing, duct insulation, the electrodes, and current leads. The costing of the other components (mixer, nozzle, etc.) were based on pressure housing materials.

The basic design used for the structural housing for all pressurized components was a reinforced (ribbed) plate construction. This allowed for minimum weight designs. The design basis for the pressurized components is given in Appendix A 11.7. The costing of structural members was based on total material weights and type. The generalized form of the equations used to calculate component weights is given in Table 11.5. The specific cost factors and the criteria used for applying these costs is given in Table 11.7.



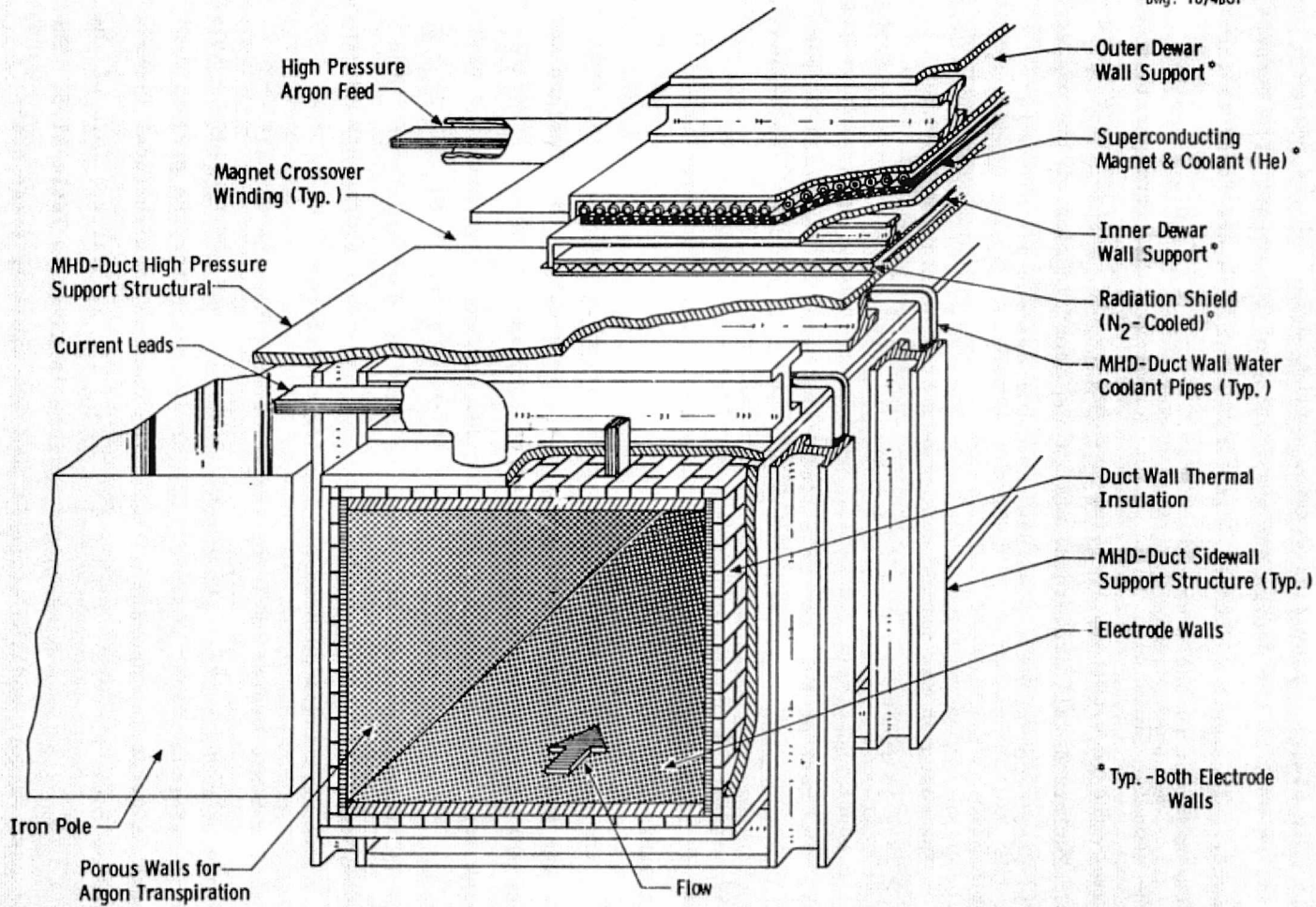


Fig. 11.8—Conceptual design for high pressure LM-MHD duct

Table 11.7 - Summary of Material Cost for Liquid-Metal MHD Components

Component	Material	Total Cost \$/lb <sup>(1)</sup>	Use Criteria
1. MHD duct	316 SS (2)	8.25	All cases
2. MHD duct installations	MgO Brick (3)	0.60	All cases
3. Mixer	316 SS	11.25	Na/A - only temp. $\leq$ 1200°F
	Haynes 25 (4)	15.00	Na/A - only temp. $>$ 1200°F
	Nb-1Zr (5)	43.80	Li/He case
4. Nozzle	316 SS	8.25	Na/A - temp. $\leq$ 1200°F
	Haynes 25	12.00	Na/A - temp. $>$ 1200°F
	Nb-1Zr	43.90	Li/He case
5. Separator/diffuser	316 SS	14.25	Na/A - temp. $\leq$ 1200°F
	Haynes 25	18.00	Na/A - temp. $>$ 1200°F
	Nb-1Zr	43.90	Li/He case
6. Liquid metal (6) piping and fittings	316 SS	8.40	Na/A - temp. $\leq$ 1200°F
	Haynes 25	22.40	Na/A - temp. $>$ 1200°F
	Nb-1Zr	123.00	Li/He case

(1) Includes material costs, fabrication and overhead as base, direct costs (no escalation or contingency)

(2) Material cost estimate - United States Steel - private communication, fabrication cost estimate - Ionic's Corp. - private communication; overhead taken as 50% of material and fabrication costs

(3) Cost estimate obtained from Harbison/Walker Refractories Co. - private communication

(4) Haynes Metal Alloy Data Sheets, #25, Union Carbide, Stellite Division.

(5) GEAP-14018 - UC-77, "VTHR for Process Heat," General Electric, September 1974.

(6) Costs based on CRBRP large seamless stainless steel pipe costs. Includes fittings, heat tracing, standard insulation, fabrication and erection. Piping cost for Haynes & Nb-1Zr obtained by direct ratio of material costs

The MHD duct insulation costs were based on the quantity of magnesium oxide brick required to line the two duct walls. The current leads from the duct were copper bus bars sized to handle approximately 3.7 MA of generator total current. The cost basis for these leads was flat copper bar at \$2.44/kg (\$1.107/lb). The total cost (including installation) of the electrode system was estimated for an MHD generator producing 1750 MWe (total) at \$4 million. For the parametric case in which the duct power differed appreciably from this design, electrode cost was scaled linearly with power output.

### 11.5.3 Liquid-Metal Pumps and Subsystem Costs

A second major set of components that required design emphasis were the liquid-metal pump and associated liquid-metal subsystems, which include the purification system and inventory and emergency dump capacities. For these systems, available liquid-metal fast breeder reactor technology was utilized to develop sizing and cost information.

The liquid-metal purification system was designed with current cold trap (oxide control) technology as developed by Mine Safety Appliance (MSA). The basic criterion for designing and sizing the system (see Appendix A 11.5) is the assumed cover gas leak rate. The value chosen for this parameter in this study was commensurate with current liquid-metal fast breeder reactor design. The algorithm developed for costing the system was based on cost estimates (made by MSA) for two different sized currently available systems.

The liquid-metal storage and emergency dump systems were sized to meet inventory requirements. The costing of the system (as given by the equation in the Table 11.5) was again based on current breeder reactor design estimates. The costing includes piping, tank, heat tracing, and valves.

### 11.5.4 Heat Transfer Equipment Costs

The major heat transfer equipment for which design and costing

had to be established were:

- Primary heat exchanger (includes a coal combustor and the radiant and convective heat exchanger).
- Steam generator
- Compressor intercooler (if used) and
- The gas to air preheaters and recuperators.

In general, to obtain an economic and optimum heat exchanger design, a trade-off must be made between tube size, pressure loss, and heat transfer coefficients. This usually involves lengthy computer solutions. The design approach used herein, however, was to make some simplifying assumptions based on engineering experience that would avoid the iterative solution. These assumptions were to consider only counter-flow shell and tube designs, treat only the controlling-side heat transfer coefficients (i.e., neglecting water side where applicable) and, based on well-established experience, a priori select both a tube geometry (diameter and pitch) and pressure drop. In all cases, this approach led to a reasonable design but one that may not be the most economical, considering the total system.

A preliminary design was prepared for both the steam generator and primary heat exchangers for the base case conditions. These designs are summarized in Appendix A 11.6. The theoretical basis for the heat exchanger designs is the same as is given in Appendix A 9.3 of the open-cycle MHD study. For the remaining parametric cases, the heat exchange equipment was modeled from these designs. In all cases, established correlations and equations were employed to calculate heat transfer coefficients, mean temperature differences, and pressure drops. With the air-to-gas (or gas-to-gas) type exchangers, the designs were based on heat exchanger effectiveness, a concept widely applied in heat exchanger design. Once surface area requirements were determined, geometry factors (tube diameter, pitch, etc.) were employed to calculate flow area requirements and, ultimately, overall physical size.

The overall cost of the heat exchange equipment was determined from both tube and shell costs. The shell costs were based on material

Table 11.8 Summary of Cost of Heat Exchanger Component Materials

Heat Exchanger Component	Material(s)	Costs, \$/lb	Use Criteria
1. Primary heat exchanger radiation tubes - 2 in id	Haynes 188-Clad 2 in i.d.	14.00 <sup>a</sup>	Flue gas temp. < 3000°F, Na/A
2. Radiation/convection tubes- 2 in id	Cb-1Zr Haynes 188	50.2 <sup>b</sup> 14.00	Li/He case Flue gas temp.< 1800°F, Na/A
3. Convection tubes - finned - 1.8 in id	316 SS	5.83	Flue gas temp.< 1200°F, Na/A
4. Steam generator/precooler convection tubes - 1.0 in od	316 SS Haynes 188 Nb-1Zr	5.48 <sup>c</sup> 12.1 50.2	Argon temp. < 1200°F Argon temp. < 1500°F Li/He case
5. Shell	Carbon steel, lined	0.67	All cases

<sup>a</sup>Cost data obtained from Reference 11.1 adjusted to 1974 dollars and corrected for cladding.

<sup>b</sup>GEAP-14018-uc-77 "VTHR for Process Heat."

<sup>c</sup>Tube material cost estimate obtained from Westinghouse, Tampa Nuclear Steam Generator Division, private communication.

type, insulation, and weight. Tube costs were broken into material and fabrication. The fabrication costs were based on the number of tube end welds required, taken at \$10 per weld. This factor is approximately correct for 2.54 cm (1 in) diameter tubes. Table 11.8 summarizes the material and fabrication cost factors used for the different heat exchanger types.

#### 11.5.5 Costs of Gas Compressors and Motor Drives

Guidelines for costing the large gas compressor were obtained from the Westinghouse Gas Turbine Division. From this information an algorithm was developed that relates costs to the required compressor flow rates and pressure ratios.

The motor drives used for the pump and/or compressor were costed at \$36 per kW (installed), consistent with 1974-75 market prices (Westinghouse Large Motor Division).

#### 11.5.6 Steam Turbine/Generator Costs

Costs for the steam plant electric generators and turbine drives (for those LM-MHD cases that used a bottoming plant) were obtained from the Westinghouse Steam Turbine Division. The costs were derived from the heat load and plant configuration that was determined to be best suited for the LM-MHD binary plant. The major steam plant costs, for the base case study point and each parametric point that involved a steam plant, were:

- Turbine/generators sets - \$22.25/kW
- Turbine drives - \$13.14/kW.

Since this plant did not incorporate extractive feedwater heating, there were no feedwater heaters involved. The boiler feed pump and auxiliary steam equipment costs were included in the balance of plant costs (see Section 2.0).

#### 11.5.7 Sizing and Cost Algorithm Computer Program

To help facilitate the LM-MHD system analysis, the above described sizing, weight, and cost algorithms for each major component were programmed.

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Table 11.9 Summary of Size, Weight and Cost Calculations for LM-MHD Cycle Components - Base Case

SYSTEM	MAJOR COMPONENTS	NUMBER OF COMPONENTS	TOTAL WEIGHT N LOOPS(LBS)	CHARACTERISTIC SIZE		COST FOR N LOOPS (DOLLARS) N = 16.
				WIDTH	LENGTH	
LIQUID METAL LOOP (SODIUM)	MHD DUCT INDUCTOR SUPPORT INSULATION	•16000+02 •16000+02	•15219+06 •11400+06	•21308+01 FT •20000+00 FT	•22775+02 FT •22775+02 FT	•12556+07 •68399+05
	2 MIXER	•16000+02	•40226+05	•20704+01 FT	•20704+01 FT	•45255+06
	3 NOZZLE	•16000+02	•25738+04	•22534+01 FT	•33576+00 FT	•21234+05
	4 SEPARATOR AND DIFFUSER	•16000+02	•19048+04	•15023+01 FT	•67151+00 FT	•28284+05
	SPRIMARY RETURN LEG PIPING	•16000+02	•19249+06	•24498+01 FT	•30407+02 FT	•16169+07
	TOTAL WEIGHT LIQUID METAL DUCTING		•50346+06	•21308+01 FT	•30407+02 FT	•34429+07
MAGNET SYSTEM	6 IRON POLES	•16000+02	•74396+06	•20704+01 FT	•22775+02 FT	•53566+06
	7 DEMAG SET	•16000+02	•33050+04	•21308+01 FT	•22775+02 FT	•27273+05
	8 SUPERCONDUCTOR	•16000+02		•10596+01 CU-FT		•48439+06
LIQUID METAL SUB-SYSTEMS	9 LIQUID METAL PUMP	•43095+01	•71106+06	1MP. DIA. x 80 INCH		•34476+08
	10 LIQUID METAL INVENTORY		•17449+06			•23904+06
	11 LIQUID METAL EMER. DUMP		•16380+04	•25200+05 GAL		•18318+07
	12 LIQUID METAL PURIFICATION			•53604+04 GPH		•44903+07
GAS LOOP ARGON	13 PRIMARY HX	•40000+01	•39951+07	•18129+02 FT	•16386+03 FT	•36364+08
	14 5TH. CONDENSATOR	•40000+01	•69958+07	•18855+02 FT	•93450+02 FT	•27778+08
	15 PREHEATER	•40000+01	•12705+07	•13353+02 FT	•12708+02 FT	•55348+07
	16 INTERCOOLER	•00000	•00000	•00000 FT	•00000 FT	•00000
	17 GAS COMPRESSOR	•16000+02	•26723+06	•72789+03 LB/SEC		•28269+07
	18 GAS PIPING		•27147+04		•22500+03 FT	•16014+05

This program reduced calculation time and, in addition, permitted evaluation of the sensitivity of several independent design parameters in terms of their impact on cost. This has allowed flexibility in establishing final physical configurations that are consistent with the imposed design restrictions, as well as permitting physical sizes that conform to acceptable commercial design practice. The computer program also provided a permanent record of the data in convenient table form. Samples of such printout sheets that summarize the results of the sizing, weight, and cost calculations for the base case and for Points 14 and 8 are shown in Tables 11.9, 11.10, and 11.11, respectively.

#### 11.5.8 Component Installation Costs

The installation costs for the major equipment were determined on the basis of the complexity of the particular piece of equipment and its capital costs. The general guidelines used are given in Table 11.12 (recommendations made by architect engineer Chas. T. Main, Inc.). Figure 11.9 shows the artist's conceptual plant arrangement prepared from plan and elevation drawings of the plant design. The plan and elevation drawings were used to estimate piping, installation, and balance of plant requirements. As Figure 11.9 illustrates, most of the plant systems have been enclosed in the protective building because of the high current requirements between the MHD duct and inverters. Figure 11.10 is the site layout prepared for the LM-MHD/steam power plant base case.

#### 11.5.9 Summary of Component Costs for LM-MHD Study Cases

In the LM-MHD ten power plants were costed to determine component capital and installation, indirect, and balance of plant costs. The detailed accounts listing, cost of electricity summary, and input-output sheets for the base case (Point 16) and for Points 8 and 14 are given as Tables 11.13 through 11.21, respectively. A relative breakdown of the major LM-MHD component cost items for the base case is also shown graphically in Figure 11.11. The costs are shown as a percentage of the total. A review of this figure shows that the actual MHD duct assembly and magnet costs are small compared to those of other system components.



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Table 11.10 Summary of Size, Weight and Cost Calculations for LM-MHD Cycle Component - Point 14

SYSTEM	MAJOR COMPONENTS	NUMBER OF COMPONENTS	TOTAL WEIGHT (N LOOPS(LBS))	CHARACTERISTIC SIZE		COST FOR N LOOPS (DOLLARS) N = 15.
				WIDTH	LENGTH	
LIQUID METAL LOOP (COMMON)	MHD SUCT	.16000+01	.18073+05	.20357+01 FT	.24470+02 FT	.14915+07
	15 STRUCT. SUPPORT INSULATION	.16000+02	.13243+06	.21000+00 FT	.26430+02 FT	.79459+05
	3 MIXER	.16000+02	.40060+05	.27209+01 FT	.22203+01 FT	.60089+06
	4 NOZZLE	.16000+02	.27592+04	.24743+01 FT	.35963+00 FT	.33230+05
	4 SEPARATOR AND DIFFUSER	.16000+02	.21967+04	.15497+01 FT	.73730+00 FT	.38641+05
	16 PRIMARY RETURN LOOP PIPING	.16000+02	.19074+06	.25273+01 FT	.32543+02 FT	.42725+07
	TOTAL WEIGHT LIQUID METAL SUCTION		.54893+06	.27857+01 FT	.32640+02 FT	.65163+07
MAGNET SYSTEM	6 IRON POLES	.15000+02	.91326+06	.27203+01 FT	.24430+02 FT	.56115+06
	7 BEAR SET	.15000+02	.34986+04	.22857+01 FT	.24430+02 FT	.28863+05
	8 SUPERCONDUCTOR	.15000+02		.99755+00 CU-FT		.45602+06
LIQUID METAL SUB-SYSTEMS	9 LIQUID METAL PUMP	.40580+01	.91817+06	IMP. DIA.=30 INCH		.39669+08
	10 LIQUID METAL INVENTORY		.21553+06			.29527+06
	11 LIQUID METAL ENTR. DUMP		.20000+04	.33773+00 GAL		.21237+07
	12 LIQUID METAL PURIFICATION			.01678+04 GPM		.48847+07
GAS LOOP ARCON	14 PRIMARY HX	.40000+01	.05650+07	.29801+02 FT	.97021+02 FT	.79436+08
	15 STEAM GENERATOR	.40000+01	.40742+07	.20220+02 FT	.44320+02 FT	.20912+08
	16 PREHEATER	.40000+01	.19755+07	.14743+02 FT	.15793+02 FT	.79420+07
	18 INTERCOOLER	.00000	.00000	.00000 FT	.00000 FT	.00000
	17 GAS COMPRESSOR	.15000+02	.26422+06	.70932+02 LB/SEC		.27743+07
	18 GAS PIPING		.02264+04		.00000+03 FT	.13626+05

Table 11.11 Summary of Size, Weight and Cost Calculations for LM-MHD Cycle Components - Point 8

SYSTEM	MAJOR COMPONENTS	NUMBER OF COMPONENTS	TOTAL WEIGHT N LOOPS (LBS)	CHARACTERISTIC SIZE		COST FOR N LOOPS (DOLLARS)
				WIDTH	LENGTH	
LIQUID METAL LOOP (500TON)	MHD DUCT					
	1 STRUCT. SUPPORT	.16000+02	.21835+06	.24660+01 FT	.26358+02 FT	.18019+07
	2 INSULATION	.16000+02	.15418+06	.21000+00 FT	.26358+02 FT	.92493+05
	3 MIXER	.16000+02	.52847+05	.23961+01 FT	.23961+01 FT	.79270+06
	4 NOZZLE	.16000+02	.37577+04	.26938+01 FT	.40137+00 FT	.45092+05
LIQUID METAL SUB-SYSTEMS	5 SEPARATOR AND DIFFUSER	.16000+02	.29319+04	.17958+01 FT	.80273+00 FT	.52774+05
	6 PRIMARY FLUID LEG PIPING	.16000+02	.24992+06	.28352+01 FT	.35229+02 FT	.65982+07
	7 TOTAL WEIGHT LIQUID METAL DUCTING		.60196+06	.24660+01 FT	.35229+02 FT	.83826+07
MAGNET SYSTEM	8 IRON POLES	.16000+02	.11532+07	.23961+01 FT	.26358+02 FT	.83032+06
	9 BEAR SET	.16000+02	.46104+04	.24660+01 FT	.26358+02 FT	.38036+05
LIQUID METAL SUB-SYSTEMS	10 SUPERCONDUCTOR	.16000+02		.11612+01 CU-FT		.53083+06
	11 LIQUID METAL PUMP	.57720+01	.95238+06	IMP. DIA.=80 INCH		.46176+08
	12 LIQUID METAL INVENTORY		.27076+06			.37094+06
	13 LIQUID METAL EMER. DUMP		.25418+04	.40669+05 GAL		.24919+07
GAS LOOP ARGON	14 LIQUID METAL PURIFICATION			.71796+04 GPM		.53508+07
	15 PRIMARY MX	.40000+01	.76428+07	.32155+02 FT	.97021+02 FT	.92478+08
	16 STM. GENERATOR	.40000+01	.47773+07	.21822+02 FT	.44249+02 FT	.24329+08
	17 PREHEATER	.40000+01	.22402+07	.15900+02 FT	.15805+02 FT	.91233+07
	18 INTERCOOLER	.00000	.00000	.00000 FT	.00000 FT	.00000
	19 GAS COMPRESSOR	.16000+02	.29561+06	.82567+03 LB/SEC		.31039+07
	20 GAS PIPING		.25941+04		.20000+03 FT	.15876+05

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Table 11.12 Summary of Basis for Estimating Installation Cost of Major LM-MHD Components

<u>Cost Basis, % of capital cost</u>	<u>Criteria</u>	<u>LM-MHD Components</u>
10	Simple component installation on foundations	<ul style="list-style-type: none"> <li>• LM pump</li> <li>• LM inventory</li> <li>• LM emergency dump</li> <li>• Compressors</li> <li>• Motors</li> <li>• Turbines</li> </ul>
20	Electrical equipment and sophisticated support components	<ul style="list-style-type: none"> <li>• LM auxiliary systems</li> <li>• Superconducting magnet</li> <li>• LM purification</li> <li>• Inverters</li> <li>• Transformers</li> <li>• Circuit breakers</li> <li>• Current conductors</li> </ul>
40	Field assembly and erection	<ul style="list-style-type: none"> <li>• MHD duct and component assembly</li> <li>• Support structure</li> <li>• LM piping</li> <li>• Gas piping</li> <li>• Steam generators</li> <li>• Primary heat exchanger</li> <li>• Air preheaters &amp; gas recuperators</li> </ul>

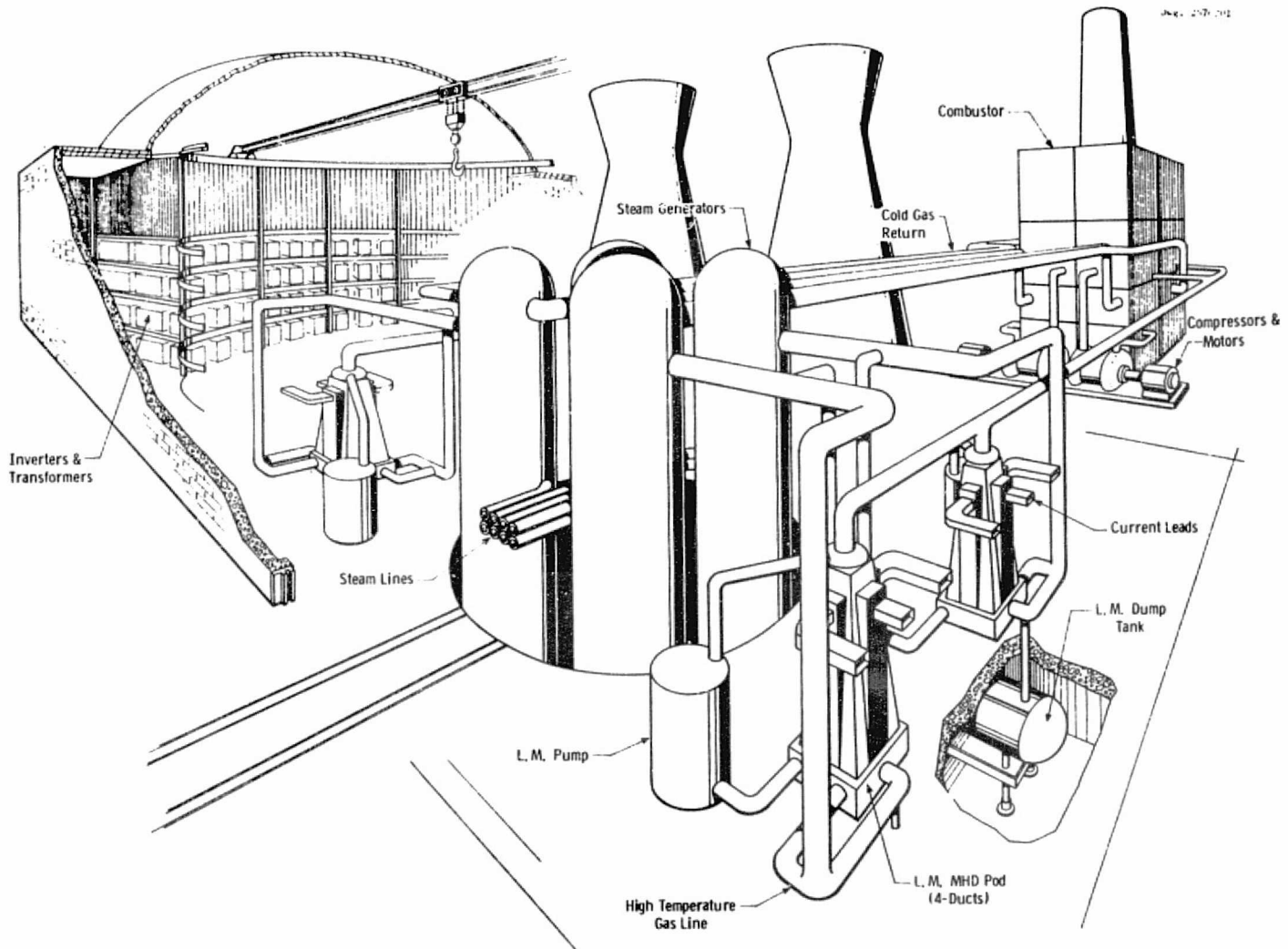


Fig. 11.9—Conceptual component arrangement for L. M. MHD/steam power plant

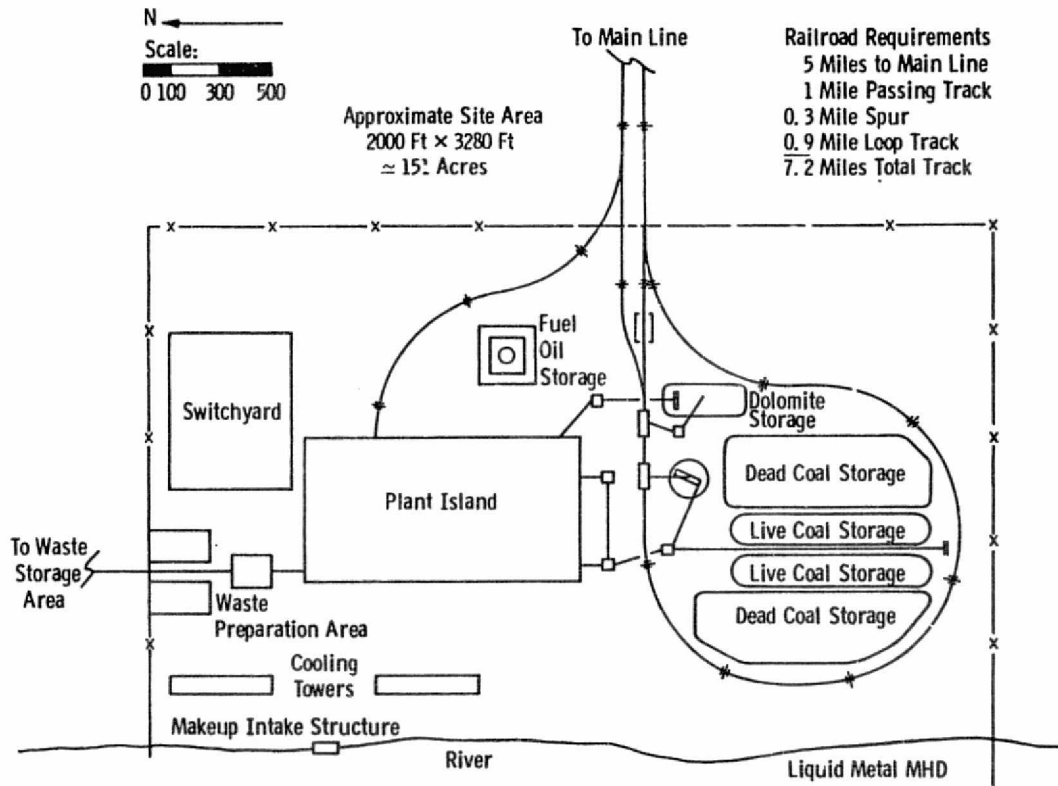


Fig. 11. 10—Plant island arrangement (LM-MHD Base Case, Point 16)

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Table 11.13 CLOSE CYCLE LIQUID METAL WAF SYSTEM ACCOUNT LISTINGS  
PARAMETRIC POINT NO. 11

ACCOUNT NO. & NAME	UNIT	AMOUNT	MAT \$/UNIT	INS \$/UNIT	MAT COST,\$	INS COST,\$
<b>SITE DEVELOPMENT</b>						
1. 1 LAND COST	ACRE	182.0	1000.00	.00	182000.00	.00
1. 2 CLEARING LAND	ACRE	57.7	.22	600.00	.00	34639.54
1. 3 GRADING LAND	ACRE	182.0	.00	3000.00	.00	546000.00
1. 4 ACCESS RAILROAD	MILE	5.1	115000.00	110000.00	575000.00	550000.00
1. 5 LOOP RAILROAD TRACK	MILE	2.5	120000.00	70000.00	300000.00	170000.00
1. 6 SIDING R R TRACK	MILE	.1	25000.00	90000.00	.00	.00
1. 7 OTHER SITE COSTS	ACRE	.0	.00	.00	387031.29	387031.29
PERCENT TOTAL DIRECT COST IN ACCOUNT 1 =			.437	ACCOUNT TOTAL,\$	1844331.29	1694427.64
<b>EXCAVATION &amp; PILING</b>						
2. 1 COMMON EXCAVATION	YDS	131816.0	.00	3.00	2355808.00	407700.00
2. 2 PILING	FT	372450.0	6.57	9.50	2355808.00	3098400.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 2 =			1.059	ACCOUNT TOTAL,\$	2355808.00	3488100.00
<b>PLANT ISLAND CONCRETE</b>						
3. 1 PLANT IS. CONCRETE	YDS	18713.7	70.00	90.00	1291300.00	1454000.00
3. 2 SPECIAL STRUCTURES	YDS	37000.0	30.00	150.00	2430000.00	4050000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 3 =			2.401	ACCOUNT TOTAL,\$	3711000.00	5514000.00
<b>HEAT REJECTION SYSTEM</b>						
4. 1 COOLING TOWERS	EACH	17.0	.00	.00	2609500.00	1300500.00
4. 2 CIRCULATING WZJ SYS	EACH	1.0	.00	.00	1431795.56	2002256.67
4. 3 SURFACE CONDENSER	FT2	502924.3	.00	.00	2729438.17	352647.46
PERCENT TOTAL DIRECT COST IN ACCOUNT 4 =			2.010	ACCOUNT TOTAL,\$	6130693.69	3652803.09
<b>STRUCTURAL FEATURES</b>						
5. 1 STAT. STRUCTURAL ST. TON		2875.0	650.00	175.00	1868750.00	503125.00
5. 2 SILOS & BUNKERS	TP4	443.1	1800.00	750.00	737513.69	332297.12
5. 3 CHIMNEY	FT	400.0	.00	.00	435070.92	852006.30
5. 4 STRUCTURAL FEATURES	EACH	1.0	1434000.00	405000.00	1434000.00	405000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 5 =			1.715	ACCOUNT TOTAL,\$	4537333.64	1857028.50
<b>BUILDINGS</b>						
6. 1 STATION BUILDINGS	FT2	734000.0	.15	.15	1174400.00	1174400.00
6. 2 ADMINISTRATION	FT2	10000.0	10.00	14.00	160000.00	140000.00
6. 3 WAREHOUSE & SHOP	FT2	20000.0	12.00	3.00	240000.00	160000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 6 =			.813	ACCOUNT TOTAL,\$	1574400.00	1474400.00
<b>FUEL HANDLING &amp; STORAGE</b>						
7. 1 COAL HANDLING SYS	TP4	443.1	.00	.00	5140501.00	2537386.62
7. 2 DOLOMITE HAND. SYS	TP4	66.0	.00	.00	1104083.72	586144.73
7. 3 FUEL OIL HAND. SYS	3AL	5250.0	.00	.00	74450.99	63272.91
PERCENT TOTAL DIRECT COST IN ACCOUNT 7 =			2.823	ACCOUNT TOTAL,\$	7244635.62	3256804.25
<b>FUEL PROCESSING</b>						
8. 1 COAL DRYER & CRUSHER	TP4	443.1	.00	.00	1519709.77	1013139.84
8. 2 CARBONIZERS	TP4	.0	.00	.00	.00	.00
8. 3 GASTIERS	TP4	.1	.00	.00	.00	.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 8 =			.076	ACCOUNT TOTAL,\$	1519709.77	1013139.84

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Table 11.13 CLOSE CYCLE LIQUID METAL MHD SYSTEM ACCOUNT LISTING  
(Continued) PARAMETRIC POINT NO.10

ACCOUNT NO. & NAME	UNIT	AMOUNT	MAT \$/UNIT	INS \$/UNIT	MAT COST,\$	INS COST,\$
<b>FIRING SYSTEM</b>						
9. 1						
PERCENT TOTAL DIRECT COST IN ACCOUNT 9 =		.00	.00	.00	.00	.00
ACCOUNT TOTAL,\$						
<b>VAPOR GENERATOR (FIRED)</b>						
10. 1						
PERCENT TOTAL DIRECT COST IN ACCOUNT 10 =		.00	.00	.00	.00	.00
ACCOUNT TOTAL,\$						
<b>ENERGY CONVERTER</b>						
11. 1	MHD DUCT STRUCT STEEL	KP	152.2	1255000.00	501000.00	1255000.00
11. 2	DUCT INSULATION	KP	114.3	85000.00	27000.00	58000.00
11. 3	MIXER	KP	40.2	452000.00	181000.00	452000.00
11. 4	NOZZLE	KP	2.5	21000.00	10000.00	21000.00
11. 5	SEPARATOR/DIFFUSER	KP	2.0	28000.00	11000.00	28000.00
11. 6	LM PIPING/FITTINGS/INS	KP	192.5	1320000.00	542000.00	1620000.00
11. 7	IRON POLES	KP	744.0	535000.00	214000.00	535000.00
11. 8	AUX EQUIPMENT & SYSTEM		1.0	.00	.00	.00
11. 9	MAGNET DEWAR	KP	3.3	30000.00	11000.00	30000.00
11.10	SUPERCON MAG COOLANT FT3		1.1	494000.00	97000.00	494000.00
11.11	ASSEMBLY SUPPORT STRUCT		1.0	.00	.00	.00
11.12	L M PUMPS	KP	711.1	7470000.00	3450000.00	3470000.00
11.13	L M INVENTORY (SODIUM)KP		174.5	24000.00	24000.00	24000.00
11.14	L M EMERGENCY DUMP	3AL	1.6	132000.00	183000.00	183000.00
11.15	L M PURIFICATION	GPM	5.4	449000.00	896000.00	449000.00
11.16	GAS COMPRESSOR	KP	253.2	233000.00	237000.00	233000.00
11.17	MOTOR DRIVES	MW	435.4	1740000.00	1800000.00	1740000.00
11.18	GAS INVENTORY	KP	185.3	29000.00	3000.00	29000.00
11.19	GAS PIPING	KP	2.7	16600.00	6700.00	16600.00
11.20	STEAM TURBINES-GEN	MW	255.0	5690000.00	5690000.00	5690000.00
11.21	DRIVE TURBINES	MW	658.0	3170000.00	3170000.00	3170000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 11 =		19.950			4399500.00	3325700.00
ACCOUNT TOTAL,\$						
<b>COUPLING HEAT EXCHANGER</b>						
12. 1	COMB HEAT EXCHANGER	KP	3996.1	36760000.00	14560000.00	36760000.00
12. 2	OPEN CYCLE MHD TOPPING		1.0	1.00	1.00	1.00
12. 3	STEAM GENERATOR	KP	6995.8	27778000.00	11120000.00	27778000.00
12. 4	INSULATION		1.0	374000.00	255000.00	374000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 12 =		25.667			67978001.00	2824001.00
ACCOUNT TOTAL,\$						
<b>HEAT RECOVERY HEAT EXCH.</b>						
13. 1	AIR PREHEATERS	KP	1273.5	5730000.00	553000.00	5730000.00
13. 2	GAS RECUPERATORS	KP	1.0	1.00	1.00	1.00
13. 3	COMP INTERCOLLER	KP	1.0	1.00	1.00	1.00
13. 4	INSULATION	KP	1.0	330000.00	220000.00	330000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 13 =		1.769			5950002.00	773002.00
ACCOUNT TOTAL,\$						
<b>WATER TREATMENT</b>						
14. 1	DEMINEALIZER	GPM	152.6	2500.00	700.00	381520.00
14. 2	CONDENSATE POLISHING	KWE	953800.0	1.25	.30	1192250.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 14 =		.525			1573765.98	352965.60
ACCOUNT TOTAL,\$						

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Table 11.13 CLOSE CYCLE LIQUID METAL M.I. SYSTEM ACCOUNT LISTINGS  
(Continued) PARAMETRIC POINT NG.12

ACCOUNT NO.	#	NAME	UNIT	AMOUNT	MAT \$/UNIT	INS \$/UNIT	MAT COST*	INS COST*
<b>POWER CONDITIONING</b>								
15.	1	INVERTERS		3972.3	52750000.00	7340000.00	72750000.00	7340000.00
15.	2	TRANSFORMERS	KP	7812.7	1550000.00	190000.00	3550000.00	1930000.00
15.	3	CIRCUIT BREAKERS		144.0	63000.00	72600.00	363000.00	72600.00
15.	4	COOLING SYSTEM		1.0	1.00	1.00	1.00	1.00
15.	5	CONDENSERS		1.0	200000.00	112000.00	200000.00	112000.00
15.	6	STD TRANSFORMERS		31230.3	.30	.00	52833.23	10575.60
PERCENT TOTAL DIRECT COST IN ACCOUNT 15 =				10.121	ACCOUNT TOTAL*		49991835.00	10443177.62
<b>AUXILIARY MECH EQUIPMENT</b>								
15.	1	BOILER FEED PUMP	934. KWC	925043.6	1.57	.13	1544931.08	92504.95
15.	2	OTHER PUMPS	KWE	29056.8	1.38	.12	256041.96	34914.81
15.	3	MISC SERVICE SYS	KWE	37885.9	1.17	.73	113471.39	70799.90
15.	4	AUXILIARY BOILER	PPH	28000.0	4.20	.80	112000.00	22400.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 15 =				1.354	ACCOUNT TOTAL*		4055604.41	1059414.45
<b>PIPE &amp; FITTINGS</b>								
17.	1	CONVENTIONAL PIPING	TON	7800.0	3000.00	1800.00	11400000.00	6840000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 17 =				4.366	ACCOUNT TOTAL*		11400000.00	6840000.00
<b>AUXILIARY ELEC EQUIPMENT</b>								
18.	1	MISC MOTORS, ETC		280556.8	1.40	.17	407339.48	49462.65
18.	2	SWITCHGEAR & MCC PAN	KWE	387342.4	1.35	.45	754497.59	174574.06
18.	3	CONDUIT, CABLES, TRAYS	FT	430000.0	1.22	1.36	5794799.54	5270399.54
18.	4	ISOLATED PHASE BUS	FT	500.0	510.00	950.00	255000.00	225000.00
18.	5	LIGHTING & COMMUN	KWE	909855.9	.35	.43	334449.56	417038.04
PERCENT TOTAL DIRECT COST IN ACCOUNT 18 =				3.838	ACCOUNT TOTAL*		7553075.56	6936474.52
<b>CONTROL, INSTRUMENTATION</b>								
19.	1	COMPUTER	EACH	1.0	100000.00	15000.00	60000.00	15000.00
19.	2	OTHER CONTROLS	EACH	1.0	100000.00	34100.00	100000.00	54100.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 19 =				.636	ACCOUNT TOTAL*		172000.00	65600.00
<b>PROCESS WASTE SYSTEMS</b>								
20.	1	BOTTOM ASH	TPH	34.2	3479371.59	989757.30	3479371.59	863767.90
20.	2	DRY ASH	TPH	9.6	17807.64	153451.91	613807.64	153451.91
20.	3	WET SLURRY	TPH	35.0	105488.73	451522.03	1806439.33	451622.08
20.	4	ONSITE DISPOSAL	ACRE	353.6	6203.67	9242.30	2144996.09	3284281.78
20.	5	SEED TREATMENT	EACH	.0	.00	.07	.00	.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 20 =				3.479	ACCOUNT TOTAL*		8094363.56	4759123.62
<b>STACK GAS CLEANING</b>								
21.	1	PRECIPITATOR	EACH	.1	3000337.31	5200024.55	.00	.00
21.	2	SCRUBBER	KWE	100000.0	21.42	.82	2141667.75	5812194.25
21.	3	MISC STEEL & JOISTS		.0	.00	.00	.00	.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 21 =				9.332	ACCOUNT TOTAL*		2141667.75	5812194.25
<b>TOTAL DIRECT COSTS*</b>							<b>27324824.00</b>	<b>111631754.00</b>

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Table 11.14 CLOSE CYCLE LIQUID METAL M43 SYSTEM COST OF ELECTRICITY, MILLS/KW.HR  
PARAMETRIC POINT NC.16

ACCOUNT	RATE, PERCENT	6.00	8.50	10.50	15.00	21.50
TOTAL DIRECT COSTS, \$	.0	337773226	374742974	374877576	417084340	479385896
INDIRECT COST, \$	11.0	29339977	41563551	51832193	73347443	135131335
PROF & OWNER COSTS, \$	8.0	20481859	3379478	2990206	33365147	38350055
CONTINGENCY COST, \$	11.0	35385754	79021723	41236533	45877077	52732426
SUB TOTAL, \$	.0	422959112	443707086	497936504	569654000	675600756
ESCALATION COST, \$	6.5	143033389	152350942	174345694	139456600	236552254
INTEREST DURING CONST, \$	10.0	188731948	206514640	222198206	254189840	301464976
TOTAL CAPITALIZATION, \$	.0	759794432	832983323	934470392	1023300440	1213617520
COST OF ELEC-CAPITAL	10.0	25.31874	27.75799	29.80696	34.10004	40.44209
COST OF ELEC-FUEL	.0	8.60943	8.60943	8.60943	8.60943	8.60943
COST OF ELEC-OP & MAIN	.0	.93712	.93712	.93712	.93712	.93712
TOTAL COST OF ELEC	.0	34.96529	37.30454	39.35351	43.64659	49.99863

ACCOUNT	RATE, PERCENT	CONTINGENCY, PERCENT	-5.00	0.00	11.00	5.00	20.00
TOTAL DIRECT COSTS, \$	.0		374877576	374877576	374877576	374877576	374877576
INDIRECT COST, \$	51.0		51832193	51832193	51832193	51832193	51832193
PROF & OWNER COSTS, \$	3.0		29990206	29990206	29990206	29990206	29990206
CONTINGENCY COST, \$	20.0		-18743878	0.0	41236533	18743878	74975514
SUB TOTAL, \$	.0		437936504	497936504	475433843	531675484	651675484
ESCALATION COST, \$	6.5		153344370	159907292	174345694	166470196	165158938
INTEREST DURING CONST, \$	10.0		135423370	203787729	222198206	212151578	237243144
TOTAL CAPITALIZATION, \$	.0		786724329	832983323	894470392	854065616	855077560
COST OF ELEC-CAPITAL	10.0		25.21643	27.33353	29.30696	29.45053	31.92661
COST OF ELEC-FUEL	.0		8.60943	8.60943	8.60943	8.60943	8.60943
COST OF ELEC-OP & MAIN	.0		.93712	.93712	.93712	.93712	.93712
TOTAL COST OF ELEC	.0		35.76302	38.88508	39.35351	38.00709	41.37316

ACCOUNT	RATE, PERCENT	ESCALATION RATE, PERCENT	6.00	8.50	8.70	10.00	.00
TOTAL DIRECT COSTS, \$	.0		374877576	374877576	374877576	374877576	374877576
INDIRECT COST, \$	51.0		51832193	51832193	51832193	51832193	51832193
PROF & OWNER COSTS, \$	8.0		29990206	29990206	29990206	29990206	29990206
CONTINGENCY COST, \$	11.0		41236533	41236533	41236533	41236533	41236533
SUB TOTAL, \$	.0		497936504	497936504	497936504	497936504	497936504
ESCALATION COST, \$	6.5		129852135	174345694	221623632	239237453	0.0
INTEREST DURING CONST, \$	10.0		209885628	222188206	235123534	253402054	173115218
TOTAL CAPITALIZATION, \$	.0		337694256	834470392	954385694	1040626016	671051720
COST OF ELEC-CAPITAL	10.0		27.91464	29.80696	31.81365	34.67729	22.36185
COST OF ELEC-FUEL	.0		8.60943	8.60943	8.60943	8.60943	8.60943
COST OF ELEC-OP & MAIN	.0		.93712	.93712	.93712	.93712	.93712
TOTAL COST OF ELEC	.0		37.46113	39.35351	41.36020	44.22394	31.90940

ACCOUNT	RATE, PERCENT	INT DURING CONST, PERCENT	6.00	8.00	10.00	12.50	15.00
TOTAL DIRECT COSTS, \$	.0		374877576	374877576	374877576	374877576	374877576
INDIRECT COST, \$	51.0		51832193	51832193	51832193	51832193	51832193
PROF & OWNER COSTS, \$	8.0		29990206	29990206	29990206	29990206	29990206
CONTINGENCY COST, \$	11.0		41236533	41236533	41236533	41236533	41236533
SUB TOTAL, \$	.0		437936504	497936504	497936504	497936504	497936504
ESCALATION COST, \$	6.5		174345694	174345694	174345694	174345694	174345694
INTEREST DURING CONST, \$	15.0		126245985	172962426	222198206	237437309	357052232
TOTAL CAPITALIZATION, \$	.0		798528072	845244592	894470392	95719496	1029334424
COST OF ELEC-CAPITAL	10.0		26.60982	27.16659	29.80696	31.93123	34.30111
COST OF ELEC-FUEL	.0		8.60943	8.60943	8.60943	8.60943	8.60943
COST OF ELEC-OP & MAIN	.0		.93712	.93712	.93712	.93712	.93712
TOTAL COST OF ELEC	.0		36.15637	37.71313	39.35351	41.52784	43.84766

Table 11.14 CLOSE CYCLE LIQUID METAL MHD SYSTEM: COST OF ELECTRICITY, MILLS/KW.HR  
(Continued) PARAMETRIC POINT NO.10

ACCOUNT	RATE, PERCENT	10.00	14.40	18.00	21.50	25.00
TOTAL DIRECT COSTS,\$	.0	374877576.	374877576.	374877576.	374877576.	374877576.
INDIRECT COSTS,\$	51.0	51832193.	51832193.	51832193.	51832193.	51832193.
PROF & OWNER COSTS,\$	8.0	29990206.	29990206.	29990206.	29990206.	29990206.
CONTINGENCY COST,\$	11.0	41236533.	41236533.	41236533.	41236533.	41236533.
SUB TOTAL,\$	.0	497936504.	497936504.	497936504.	497936504.	497936504.
ESCALATION COST,\$	5.5	174345594.	174345594.	174345594.	174345594.	174345594.
INTEREST DURING CONST,\$	10.0	222188206.	222188206.	222188206.	222188206.	222188206.
TOTAL CAPITALIZATION,\$	.0	894470392.	894470392.	894470392.	894470392.	894470392.
COST OF ELEC-CAPITAL	20.0	16.55842	23.84157	29.80696	35.76835	41.38856
COST OF ELEC-FUEL	.0	8.60943	8.60943	8.60943	8.60943	8.60943
COST OF ELEC-OP & MAINT	.0	.93712	.93712	.93712	.93712	.93712
TOTAL COST OF ELEC	.0	26.11597	33.39212	39.35351	45.31490	50.94510

ACCOUNT	RATE, PERCENT	.50	.85	1.50	2.50	1.02
TOTAL DIRECT COSTS,\$	.0	374877576.	374877576.	374877576.	374877576.	374877576.
INDIRECT COSTS,\$	51.0	51832193.	51832193.	51832193.	51832193.	51832193.
PROF & OWNER COSTS,\$	8.0	29990206.	29990206.	29990206.	29990206.	29990206.
CONTINGENCY COST,\$	11.0	41236533.	41236533.	41236533.	41236533.	41236533.
SUB TOTAL,\$	.0	497936504.	497936504.	497936504.	497936504.	497936504.
ESCALATION COST,\$	5.5	174345594.	174345594.	174345594.	174345594.	174345594.
INTEREST DURING CONST,\$	10.0	222188206.	222188206.	222188206.	222188206.	222188206.
TOTAL CAPITALIZATION,\$	.0	894470392.	894470392.	894470392.	894470392.	894470392.
COST OF ELEC-CAPITAL	13.0	29.80696	29.80696	29.80696	29.80696	29.80696
COST OF ELEC-FUEL	.0	5.06437	8.60943	15.19311	25.32186	10.33132
COST OF ELEC-OP & MAINT	.0	.93712	.93712	.93712	.93712	.93712
TOTAL COST OF ELEC	.0	35.60845	39.35351	45.93719	56.06593	41.07539

ACCOUNT	RATE, PERCENT	12.00	45.00	50.00	55.00	80.00
TOTAL DIRECT COSTS,\$	.0	374877576.	374877576.	374877576.	374877576.	374877576.
INDIRECT COSTS,\$	51.0	51832193.	51832193.	51832193.	51832193.	51832193.
PROF & OWNER COSTS,\$	8.0	29990206.	29990206.	29990206.	29990206.	29990206.
CONTINGENCY COST,\$	11.0	41236533.	41236533.	41236533.	41236533.	41236533.
SUB TOTAL,\$	.0	497936504.	497936504.	497936504.	497936504.	497936504.
ESCALATION COST,\$	5.5	174345594.	174345594.	174345594.	174345594.	174345594.
INTEREST DURING CONST,\$	10.0	222188206.	222188206.	222188206.	222188206.	222188206.
TOTAL CAPITALIZATION,\$	.0	894470392.	894470392.	894470392.	894470392.	894470392.
COST OF ELEC-CAPITAL	10.0	16.14943	43.05450	38.74905	29.80696	24.21816
COST OF ELEC-FUEL	.0	8.60943	8.60943	8.60943	8.60943	8.60943
COST OF ELEC-OP & MAINT	.0	.93712	.93712	.93712	.93712	.93712
TOTAL COST OF ELEC	.0	171.01033	52.60105	48.29560	39.35351	33.76470

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Table 11.15 CLOSE CYCLE LIQUID METAL MHD SYSTEM

ACCOUNT NO	AUX POWER, MWE	PERC PLANT POW	OPERATION COST	MAINTENANCE COST
4	11.77790	1.24155	72.59867	16.75256
7	4.39109	.51453	353.13753	.00000
8	.57598	.08072	.00000	.00000
14	.00000	.00000	15.50640	.00000
18	6.53600	.00000	.00000	.00000
20	17.59497	1.11579	.00000	.00000
21	17.00000	1.79703	.00000	.00000
TOTALS	51.35583	5.41361	333.98706	15.75256

CLOSE CYCLE LIQUID METAL MHD SYSTEM		BASE CASE INPUT	
NOMINAL POWER, MWE	1000.0000	NET POWER, MWE	943.5441
NOM HEAT RATE, BTU/KW-HR	5608.5725	NET HEAT RATE, BTU/KW-HR	10128.7429
OFF DESIGN HEAT RATE CONDENSER	.0000		
DESIGN PRESSURE, IN 13 4	3.5000	NUMBER OF SHELLS	2.0000
NUMBER OF TUBES/SHELL	13819.0457	TUBE LENGTH, FT	69.5667
U, BTU/HR-FT <sup>2</sup> -F	509.9535	TERMINAL TEMP DIFF, F	5.0000
HEAT REJECTION			
DESIGN TEMP, F	77.0000	APPROACH, F	15.6713
RANGE, F	23.0000	OFF DESIGN TEMP, F	51.4600
OFF DESIGN PRES, IN 13 4	.0000	LP TURBINE BLADE LEN, IN	28.5000

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1	1000.000	2	.000	3	.357	4	.000	5	8.000
5	256.000	7	3.500	9	1333.000	11	2.000	13	2.000
11	1.000	12	697.800	13	1.000	14	.000	15	.000
16	2.000	17	192.000	18	3.000	19	5.000	20	2.500
21	.000	22	18300.000	23	27000.000	24	2875.000	25	400.000
25	7340000.000	27	10000.000	28	20000.000	29	575000.000	30	.300
31	1.000	32	3800.000	33	1.000	34	.300	35	.400
36	4390000.000	37	500.000	38	1.000	39	1.000	40	1434000.000
41	4050000.000	42	660000.000	43	15000.000	44	1008000.000	45	641000.000
46	5.000	47	5.350	48	3.000	49	2.000	50	2.000
51	1.000	52	5.350						
6	152.133	7	114.000	8	47.225	9	2.574	10	1.985
11	192.490	12	743.980	13	1.000	14	3.306	15	1.060
16	1.000	17	711.060	18	174.490	19	1.638	20	5.360
21	269.200	22	485.400	23	185.000	24	2.710	25	256.000
26	639.000	27	1255000.000	28	501000.000	29	53000.000	30	27000.000
31	452000.000	32	181000.000	33	21000.000	34	10000.000	35	28000.000
36	11000.000	37	1520000.000	38	642000.000	39	535000.000	40	214000.000
41	7000.000	42	.000	43	30000.000	44	11000.000	45	484000.000
46	240000.000	47	.000	48	.000	49	34470000.000	50	3450000.000
51	895000.000	52	24000.000	53	1832000.000	54	183000.000	55	4490000.000
56	28000.000	57	2930000.000	58	283000.000	59	1740000.000	60	1800000.000
61	569000.000	62	3000.000	63	166000.000	64	6700.000	65	5690000.000
66	6995.800	67	1170000.000	68	917000.000	69	3336.100	70	1.000
71	1.000	72	1.000	73	36360000.000	74	14580000.000	75	1.000
76	121.500	77	2777000.000	78	11120000.000	79	3340000.000	80	2560000.000
81	553000.000	82	1.000	83	1.000	84	1.000	85	5530000.000
86	330000.000	87	1.000	88	1.000	89	1.000	90	1.000
91	1.000	92	220000.000	93	3978.800	94	3412.000	95	144.000
96	1900000.000	97	363000.000	98	36750000.000	99	7340000.000	100	3550000.000
101	2800000.000	102	1120000.000	103	72600.000	104	1.000	105	1.000
106	.000	107	1.000	108	1.000	109	230000.000	110	1.200
111	27000.000	112	90.000	113	150.000	114	.000	115	1.000

Table 11.16 CLOSE CYCLE LIQUID METAL FURNACE SYSTEM ACCOUNT LISTING  
PARAMETRIC POINT NO. 3

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ACCOUNT NO.	NAME	UNIT	AMOUNT	MAT \$/UNIT	INS \$/UNIT	MAT COST*	INS COST*	
<b>SITE DEVELOPMENT</b>								
1. 1	LAND COST	ACRE	133.7	1000.00	.00	133700.00	.00	
1. 2	CLEARING LAND	ACRE	10.7	.00	300.00	.00	3213.00	
1. 3	GRADING LAND	ACRE	133.7	.00	3000.00	.00	401100.00	
1. 4	ACCESS RAILROAD	MI	1.0	110000.00	110000.00	110000.00	550000.00	
1. 5	LOOP RAILROAD TRACK	MI	2.0	120000.00	70000.00	300000.00	175000.00	
1. 6	SIDING R R TRACK	MI	.5	120000.00	80000.00	60000.00	40000.00	
1. 7	OTHER SITE COSTS	ACRE	1.0	.00	.00	387731.29	387731.29	
PERCENT TOTAL DIRECT COST IN ACCOUNT 1 =						.482	ACCOUNT TOTAL*	1694427.84
<b>EXCAVATION &amp; PILING</b>								
2. 1	COMMON EXCAVATION	YD3	135957.0	.37	3.00	235560.00	407700.00	
2. 2	PILING	FT	36240.0	6.50	6.50	235560.00	308400.00	
PERCENT TOTAL DIRECT COST IN ACCOUNT 2 =						1.213	ACCOUNT TOTAL*	3498100.00
<b>PLANT ISLAND CONCRETE</b>								
3. 1	PLANT IS. CONCRETE	YD3	18700.0	70.00	80.00	1281000.00	1464000.00	
3. 2	SPECIAL STRUCTURES	YD3	27900.0	90.00	150.00	2430000.00	4050000.00	
PERCENT TOTAL DIRECT COST IN ACCOUNT 3 =						1.215	ACCOUNT TOTAL*	5514000.00
<b>HEAT REJECTION SYSTEM</b>								
4. 1	COOLING TOWERS	EACH	13.0	.00	.00	2753500.00	1377300.00	
4. 2	CIRCULATING H2O SYS	EACH	1.0	.00	.00	1618835.27	2177653.50	
4. 3	SURFACE CONDENSER	FT2	545753.1	.00	.00	2179449.44	392037.69	
PERCENT TOTAL DIRECT COST IN ACCOUNT 4 =						2.178	ACCOUNT TOTAL*	3929691.19
<b>STRUCTURAL FEATURES</b>								
5. 1	SIAT. STRUCTURAL ST. ION	TON	2875.0	550.00	175.00	1939750.00	503125.00	
5. 2	SILOS & BUNKERS	TPH	415.0	1800.00	750.00	755204.70	314668.62	
5. 3	CHIMNEY	FT	400.0	.00	.00	435970.32	652606.38	
5. 4	STRUCTURAL FEATURES EACH	1.0	1434000.00	40000.00	40000.00	1434000.00	405000.00	
PERCENT TOTAL DIRECT COST IN ACCOUNT 5 =						1.322	ACCOUNT TOTAL*	1975400.00
<b>BUILDINGS</b>								
6. 1	STATION BUILDINGS	FT3	734000.0	.15	.15	1174400.00	1174400.00	
6. 2	ADMINISTRATION	FT2	12000.0	16.00	14.00	150000.00	140000.00	
6. 3	WAREHOUSE & SHOP	FT2	20000.0	12.00	8.00	240000.00	160000.00	
PERCENT TOTAL DIRECT COST IN ACCOUNT 6 =						.533	ACCOUNT TOTAL*	1474400.00
<b>FUEL HANDLING &amp; STORAGE</b>								
7. 1	COAL HANDLING SYS	TPH	415.0	.00	.00	5840207.02	2490207.91	
7. 2	DOLOMITE HAND. SYS	TPH	52.5	.00	.00	1051151.52	571545.39	
7. 3	FUEL OIL HAND. SYS	CAL	57800.0	.00	.00	78460.98	53272.91	
PERCENT TOTAL DIRECT COST IN ACCOUNT 7 =						2.097	ACCOUNT TOTAL*	3125926.19
<b>FUEL PROCESSING</b>								
8. 1	COAL DRYER & CRUSHER	TPH	415.0	.00	.00	1462015.16	975210.11	
8. 2	CARBONIZERS	TPH	.0	.00	.00	.00	.00	
8. 3	GASIFIERS	TPH	.0	.00	.00	.00	.00	
PERCENT TOTAL DIRECT COST IN ACCOUNT 8 =						.506	ACCOUNT TOTAL*	1452315.16

Table 11.16  
(Continued)

CLOSE CYCLE LIQUID METAL MHD SYSTEM ACCOUNT LISTING  
PARAMPTIC COSENT NO.

ACCOUNT NO. & NAME	UNIT	AMOUNT	MAT \$/UNIT	INS \$/UNIT	MAT COST:\$	INS COST:\$
<b>FIRING SYSTEM</b>						
9. 1 PERCENT TOTAL DIRECT COST IN ACCOUNT 9 =		1.37			.07	.00
					.00	.00
<b>VAPOR GENERATOR (FIRED)</b>						
10. 1 PERCENT TOTAL DIRECT COST IN ACCOUNT 10 =		1.37			.30	.00
					.00	.00
<b>ENERGY CONVERTER</b>						
11. 1 MHD DUCT STRUCT STEEL	KP	218.4	1071433.33	720000.00	1331473.33	720000.00
11. 2 DUCT INSULATION	KP	164.2	92423.00	37000.00	62493.00	37000.00
11. 3 MIXER	KP	52.4	792700.00	31000.00	732700.00	31000.00
11. 4 NOZZLE	KP	3.0	48092.00	18100.00	45092.00	18100.00
11. 5 SEPARATOR/DIFFUSER	KP	2.9	52774.33	21100.00	52774.33	21100.00
11. 6 LM PIPING/FITTING/INS	KP	245.4	5798200.00	2230000.00	5598200.00	2230000.00
11. 7 IRON POLYS	KP	1153.2	33320.00	33320.00	33320.00	33320.00
11. 8 AUX EQUIPMENT & SYSTEM		1.0	.00	.00	.00	.00
11. 9 MAGNET DEWAR	KP	4.5	33036.00	7500.00	33036.00	7500.00
11.10 SUPERCON MAG COOLANT FT3		1.2	106166.00	106166.00	536930.00	106166.00
11.11 ASSEMBLY SUPPORT STRUCT		1.0	.00	.00	.00	.00
11.12 L H PUMPS	KP	952.4	46176000.00	46176000.00	46176000.00	46176000.00
11.13 L H INVENTORY (SODIUM)KIP		271.0	70940.00	37100.00	370940.00	37100.00
11.14 L H EMERGENCY DUMP	GAL	2541.0	245100.00	245100.00	245100.00	245100.00
11.15 L H PURIFICATION	SPH	7173.6	535000.00	1070000.00	535000.00	1070000.00
11.16 GAS COMPRESSOR	KP	295.6	310390.00	310390.00	310390.00	310390.00
11.17 MOTOR DRIVES	MW	1.0	3391720.00	2300000.00	3391720.00	2300000.00
11.18 GAS INVENTORY	KI	278.0	4000.00	4000.00	4000.00	4000.00
11.19 GAS PIPING	KP	2.5	15876.00	6350.00	15876.00	6350.00
11.20 STEAM TURBINE-GEN	MW	1.0	.00	.00	.00	.00
11.21 DRIVE TURBINES	MW	953.7	11754500.00	11354000.00	11354000.00	11354000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 11 =		24.66			102088491.00	13522306.00
<b>COUPLING HEAT EXCHANGER</b>						
12. 1 COMB HEAT EXCHANGER	KP	7542.3	22479000.00	9248000.00	32479000.00	9248000.00
12. 2 OPEN CYCLE MHD TOPPING		1.0	.00	.00	.00	.00
12. 3 STEAM GENERATOR	KP	4777.3	24230000.00	24320000.00	24230000.00	24320000.00
12. 4 INSULATION		1.0	700000.00	467000.00	700000.00	467000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 12 =		23.19			123307000.00	16351000.00
<b>HEAT RECOVERY HEAT EXCH.</b>						
13. 1 AIR PREHEATERS	KP	2240.2	9123300.00	912330.00	9123300.00	912330.00
13. 2 GAS SUPERHEATERS	KP	1.0	.00	.00	.00	.00
13. 3 COMP INTERCOLLER	KP	1.0	.00	.00	.00	.00
13. 4 INSULATION	KP	1.3	54300.00	36400.00	54300.00	36400.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 13 =		2.73			9671600.00	1277130.00
<b>WATER TREATMENT</b>						
14. 1 DEMINERALIZER	SPH	133.2	2500.00	700.00	340000.00	97440.00
14. 2 CONDENSATE POLISHING	KWE	37000.0	1.25	.30	1087500.00	261000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 14 =		.372			1435499.98	350440.00

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Table 11.16 (1997) CYCLE LIQUID METAL VAPOR SYSTEM ACCOUNT LISTING  
(Continued)

ACCOUNT NO. & NAME	UNIT	AMOUNT	MAT %/UNIT	INS %/UNIT	MAT COST*\$	INS COST*\$
<b>POWER CONDITIONING</b>						
15. 1 INVERTERS		5332.7	4.1	370000.00	4950000.00	3920000.00
15. 2 TRANSFORMERS	KP	5511.0	11.8	7000.00	11387000.00	2379000.00
15. 3 CIRCUIT BREAKERS		172.0	4.3	3000.00	434000.00	96800.00
15. 4 COOLING SYSTEM		1.0	.00	.00	.00	.00
15. 5 CONDUCTORS		1.0	2.0	3000.00	1130000.00	1130000.00
15. 6 STD TRANSFORMER		.0	.00	.00	528892.31	10577.85
PERCENT TOTAL DIRECT COST IN ACCOUNT 15 =		16.746	ACCOUNT TOTAL*\$		58239992.00	13515377.75
<b>AUXILIARY MISC EQUIPMENT</b>						
16. 1 BOILER FEED PUMP CDR	KWE	343774.6	1.07	.10	1400103.62	64377.46
16. 2 OTHER PUMPS	KWE	297958.3	.32	.12	255041.36	38914.81
16. 3 MISC SERVICE SYS	KWE	366825.9	1.17	.73	1134731.39	707994.80
16. 4 AUXILIARY BOILER	PP4	230000.0	0.33	.50	1120000.00	224000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 16 =		1.032	ACCOUNT TOTAL*\$		5819670.97	1051267.05
<b>PIPE &amp; FITTINGS</b>						
17. 1 CONVENTIONAL PIPING	TON	3300.0	3000.00	1300.00	11470000.00	6840000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 17 =		3.737	ACCOUNT TOTAL*\$		11400000.00	6840000.00
<b>AUXILIARY ELEC EQUIPMENT</b>						
18. 1 MISC MOTORS, ETC		200000.0	1.40	.17	407339.48	49462.85
18. 2 SWITCHGEAR & MCC PAK	KWE	357942.4	1.55	.45	756487.50	174574.00
18. 3 CONDUIT, CABLES, TRAYS	FT	430000.0	1.32	1.36	5734793.94	5970399.94
18. 4 ISOLATED PHASE BUS	FT	510.0	510.00	450.00	250000.00	225000.00
18. 5 LIGHTING & COMMUN	KWE	303355.0	.35	.43	339449.56	417338.04
PERCENT TOTAL DIRECT COST IN ACCOUNT 18 =		7.998	ACCOUNT TOTAL*\$		7513070.56	6836474.62
<b>CONTROL, INSTRUMENTATION</b>						
19. 1 COMPUTER	EACH	1.0	10000.00	15000.00	550000.00	15000.00
19. 2 OTHER CONTROLS	EACH	1.0	100000.00	541000.00	1028000.00	641000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 19 =		.495	ACCOUNT TOTAL*\$		1728000.00	656000.00
<b>PROCESS WASTE SYSTEMS</b>						
20. 1 BOTTOM ASH	TPH	52.0	211825.59	327950.40	5311025.55	827550.40
20. 2 DRY ASH	TPH	2.1	20343.20	145035.32	328343.20	145035.32
20. 3 WEI SLURRY	TPH	124.0	1726780.00	431025.00	1726780.00	431025.00
20. 4 ONSITE DISPOSAL	ACRE	374.0	9250.00	353.55	2059311.42	3130214.87
20. 5 SEED TREATMENT	EACH	.0	.00	.00	.00	.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 20 =		2.543	ACCOUNT TOTAL*\$		7714750.25	4534952.06
<b>STACK GAS CLEANING</b>						
21. 1 PRECIPITATOR	EACH	1.0	7167027.00	4983567.50	.00	.00
21. 2 SCRUBBER	KWE	130000.0	21.51	4.91	1012213.25	3901948.75
21. 3 MISC STEEL & DUCTS		.0	.00	.00	.00	.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 21 =		5.544	ACCOUNT TOTAL*\$		21512213.25	3901948.75
TOTAL DIRECT COSTS*\$					734719350.00	36329068.00

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Table 11.17 CLOSE CYCLE LIQUID METAL MHD SYSTEM COST OF ELECTRICITY, MILLS/KW.HR  
PARAMETRIC POINT NO. C

ACCOUNT	RATE, PERCENT	6.00	7.50	LABOR RATE, \$/HR 10.00	15.00	21.50
TOTAL DIRECT COSTS,\$	.0	432844375.	429455774.	431643437.	521833140.	531320776.
INDIRECT COSTS,\$	11.0	278814101.	296473720.	494733223.	62955524.	100266718.
PROF & OWNER COSTS,\$	8.0	35156791.	35995891.	33671374.	41750651.	46595662.
CONTINGENCY COSTS,\$	11.0	48354336.	50669005.	52981327.	57407145.	63945205.
SUB TOTAL,\$	.0	551387489.	579950472.	622595440.	730994466.	792038424.
ESCALATION COSTS,\$	6.5	19295768.	216503174.	217933324.	241042310.	277321480.
INTEREST DURING CONST. \$	10.0	245815382.	27946433.	277313250.	309334129.	353421760.
TOTAL CAPITALIZATION,\$	.0	380343266.	107976008.	1113432015.	1241270980.	1422781664.
COST OF ELEC-CAPITAL	13.0	32.810466	35.160477	37.21628	41.29924	47.33727
COST OF ELEC-FUEL	.0	8.09826	8.09826	8.09826	8.09826	8.09826
COST OF ELEC-OP & MAINT	.0	.92090	.92090	.92090	.92090	.92090
TOTAL COST OF ELEC	.0	41.855566	44.27437	46.22944	50.31740	56.35643

ACCOUNT	RATE, PERCENT	5.00	7.00	CONTINGENCY, PERCENT 11.00	15.00	20.00
TOTAL DIRECT COSTS,\$	.0	461648432.	41648432.	461648432.	41648432.	461648432.
INDIRECT COSTS,\$	11.0	49433823.	49433823.	49433823.	49433823.	49433823.
PROF & OWNER COSTS,\$	8.0	38531874.	38531874.	38531874.	38531874.	38531874.
CONTINGENCY COSTS,\$	23.0	24382421.	24382421.	24382421.	24382421.	24382421.
SUB TOTAL,\$	.0	545531704.	506141704.	622595440.	593596336.	655943906.
ESCALATION COSTS,\$	6.5	131210505.	13942528.	217933324.	207874756.	233171164.
INTEREST DURING CONST. \$	10.0	243426032.	254172044.	277813260.	264918050.	297156076.
TOTAL CAPITALIZATION,\$	.0	979363240.	107976008.	1113432015.	1056439344.	136271024.
COST OF ELEC-CAPITAL	18.0	32.810466	34.04277	37.21628	35.48305	39.80105
COST OF ELEC-FUEL	.0	8.09826	8.09826	8.09826	8.09826	8.09826
COST OF ELEC-OP & MAINT	.0	.92090	.92090	.92090	.92090	.92090
TOTAL COST OF ELEC	.0	41.82332	43.26294	46.22944	44.50225	48.82021

ACCOUNT	RATE, PERCENT	5.00	7.50	ESCALATION RATE, PERCENT 8.00	10.00	15.00
TOTAL DIRECT COSTS,\$	.0	431648432.	41648432.	431648432.	431648432.	431648432.
INDIRECT COSTS,\$	11.0	49433823.	49433823.	49433823.	49433823.	49433823.
PROF & OWNER COSTS,\$	8.0	38531874.	38531874.	38531874.	38531874.	38531874.
CONTINGENCY COSTS,\$	11.0	52981327.	52981327.	52981327.	52981327.	52981327.
SUB TOTAL,\$	.0	622595440.	506141704.	622595440.	622595440.	622595440.
ESCALATION COSTS,\$	6.5	162373256.	217933324.	277113596.	361710836.	461648432.
INTEREST DURING CONST. \$	10.0	262430720.	277813253.	293336360.	316041529.	216454798.
TOTAL CAPITALIZATION,\$	.0	1047389416.	118402016.	1193695934.	1301147856.	83050232.
COST OF ELEC-CAPITAL	13.0	34.34795	37.21628	37.71533	43.29047	27.31598
COST OF ELEC-FUEL	.0	8.09826	8.09826	8.09826	8.09826	8.09826
COST OF ELEC-OP & MAINT	.0	.92090	.92090	.92090	.92090	.92090
TOTAL COST OF ELEC	.0	43.86712	46.22944	46.73454	52.30956	36.93514

ACCOUNT	RATE, PERCENT	8.00	10.00	INT DURING CONST. PERCENT 10.00	12.50	15.00
TOTAL DIRECT COSTS,\$	.0	481648432.	41648432.	481648432.	481648432.	481648432.
INDIRECT COSTS,\$	11.0	49433823.	49433823.	49433823.	49433823.	49433823.
PROF & OWNER COSTS,\$	8.0	38531874.	38531874.	38531874.	38531874.	38531874.
CONTINGENCY COSTS,\$	11.0	52981327.	52981327.	52981327.	52981327.	52981327.
SUB TOTAL,\$	.0	622595440.	506141704.	622595440.	622595440.	622595440.
ESCALATION COSTS,\$	6.5	217933324.	217933324.	217933324.	217933324.	217933324.
INTEREST DURING CONST. \$	15.0	157851676.	216263770.	277813260.	359397548.	446440640.
TOTAL CAPITALIZATION,\$	.0	99942432.	107976008.	1113432015.	1199936304.	1237029392.
COST OF ELEC-CAPITAL	18.0	33.21904	35.160477	37.21628	39.92466	42.82067
COST OF ELEC-FUEL	.0	8.09826	8.09826	8.09826	8.09826	8.09826
COST OF ELEC-OP & MAINT	.0	.92090	.92090	.92090	.92090	.92090
TOTAL COST OF ELEC	.0	42.23820	44.18153	46.22944	48.94387	51.93983

Table 11.17 CLOSE CYCLE LIQUID METAL MHD SYSTEM COST OF ELECTRICITY, MILLS/KW.HR  
(Continued) PARAMETRIC POINT NO. 1

ACCOUNT	RATE, PERCENT	10.00	14.40	18.00	21.00	25.00
TOTAL DIRECT COSTS,\$	1.0	431549432	41648432	431549432	431549432	431549432
INDIRECT COSTS,\$	1.0	42433823	42433823	42433823	42433823	42433823
PROF & OWNER COSTS,\$	3.0	33531874	33531874	33531874	33531874	33531874
CONTINGENCY COST,\$	11.0	52981327	52981327	52981327	52981327	52981327
SUB TOTAL,\$	6.0	622595440	622595440	622595440	622595440	622595440
ESCALATION COST,\$	5.5	217993324	217993324	217993324	217993324	217993324
INTEREST DURING CONST,\$	10.0	277813260	277813260	277813260	277813260	277813260
TOTAL CAPITALIZATION,\$	10.0	1118402016	1118402016	1118402016	1118402016	1118402016
COST OF ELEC-CAPITAL	25.0	37.21023	37.21023	37.21023	37.21023	37.21023
COST OF ELEC-FUEL	0.0	8.09826	8.09826	8.09826	8.09826	8.09826
COST OF ELEC-OP & MAIN	0.0	.92090	.92090	.92090	.92090	.92090
TOTAL COST OF ELEC	0.0	29.69154	29.69154	29.69154	29.69154	29.69154

ACCOUNT	RATE, PERCENT	10.00	14.40	18.00	21.00	25.00
TOTAL DIRECT COSTS,\$	1.0	431549432	41648432	431549432	431549432	431549432
INDIRECT COSTS,\$	1.0	42433823	42433823	42433823	42433823	42433823
PROF & OWNER COSTS,\$	3.0	33531874	33531874	33531874	33531874	33531874
CONTINGENCY COST,\$	11.0	52981327	52981327	52981327	52981327	52981327
SUB TOTAL,\$	6.0	622595440	622595440	622595440	622595440	622595440
ESCALATION COST,\$	5.5	217993324	217993324	217993324	217993324	217993324
INTEREST DURING CONST,\$	10.0	277813260	277813260	277813260	277813260	277813260
TOTAL CAPITALIZATION,\$	10.0	1118402016	1118402016	1118402016	1118402016	1118402016
COST OF ELEC-CAPITAL	18.0	37.21023	37.21023	37.21023	37.21023	37.21023
COST OF ELEC-FUEL	0.0	4.75383	4.75383	4.75383	4.75383	4.75383
COST OF ELEC-OP & MAIN	0.0	.92090	.92090	.92090	.92090	.92090
TOTAL COST OF ELEC	0.0	42.88496	42.88496	42.88496	42.88496	42.88496

ACCOUNT	RATE, PERCENT	12.00	45.00	50.00	65.00	80.00
TOTAL DIRECT COSTS,\$	1.0	431549432	41648432	431549432	431549432	431549432
INDIRECT COSTS,\$	1.0	42433823	42433823	42433823	42433823	42433823
PROF & OWNER COSTS,\$	3.0	33531874	33531874	33531874	33531874	33531874
CONTINGENCY COST,\$	11.0	52981327	52981327	52981327	52981327	52981327
SUB TOTAL,\$	6.0	622595440	622595440	622595440	622595440	622595440
ESCALATION COST,\$	5.5	217993324	217993324	217993324	217993324	217993324
INTEREST DURING CONST,\$	10.0	277813260	277813260	277813260	277813260	277813260
TOTAL CAPITALIZATION,\$	10.0	1118402016	1118402016	1118402016	1118402016	1118402016
COST OF ELEC-CAPITAL	13.0	37.21023	37.21023	37.21023	37.21023	37.21023
COST OF ELEC-FUEL	0.0	8.09826	8.09826	8.09826	8.09826	8.09826
COST OF ELEC-OP & MAIN	0.0	.92090	.92090	.92090	.92090	.92090
TOTAL COST OF ELEC	0.0	210.57484	210.57484	210.57484	210.57484	210.57484

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Table 11.18

CLOSE CYCLE LIQUID METAL MHD SYSTEM

ACCOUNT NO	AUX POWER, MWE	PERC PLANT TOU	OPERATION COST	MAINTENANCE COST
4	12.77955	1.33754	79.73321	17.35490
7	4.60160	.42430	334.45079	.00000
9	.54543	.05740	.00000	.00000
14	.00000	.00000	15.00000	.00000
18	5.00000	.50000	.00000	.00000
20	9.99699	1.00000	.00000	.00000
21	17.29723	1.73370	.00000	.00000
TOTALS	49.85456	5.24708	274.99462	17.35490

CLOSE CYCLE LIQUID METAL MHD SYSTEM		CASE CASE INPUT	
NOMINAL POWER, MWE	1000.0000	NET POWER, MWE	950.1454
NOM HEAT RATE, BTU/KW-HR	9752.3353	NET HEAT RATE, BTU/KW-HR	9527.3694
OFF DESIGN HEAT RATE	.0000		
CONDENSER			
DESIGN PRESSURE, IN HG A	3.5500	NUMBER OF SHELLS	2.0000
NUMBER OF TUBES/SHELL	14395.2629	TUBE LENGTH, FT	53.0000
U, BTU/HR-FT <sup>2</sup> -F	608.8535	TERMINAL TEMP DIFF, F	5.0000
HEAT REJECTION			
DESIGN TEMP, F	77.0000	APPROACH, F	15.6713
RANGE, F	23.0000	OFF DESIGN TEMP, F	51.4000
OFF DESIGN PRES, IN HG A	.0000	LP TUBINE BLADE LEN, IN	28.5000

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1	1000.000	2	.000	3	.377	4	.530	5	3.397
6	.000	7	3.000	8	1414.000	9	2.000	10	2.000
11	1.000	12	370.500	13	1.000	14	.000	15	1.000
16	2.000	17	182.000	18	3.000	19	5.000	20	2.500
21	.000	22	1330.000	23	27000.000	24	2375.000	25	400.000
26	734000.000	27	16000.000	28	20000.000	29	575000.000	30	.300
31	1.000	32	3300.000	33	1.000	34	.300	35	.400
36	439000.000	37	500.000	38	1.000	39	1.000	40	1434000.000
41	405000.000	42	55000.000	43	1500.000	44	100000.000	45	541000.000
46	6.000	47	.000	48	2.000	49	2.000	50	2.000
51	.500	52	5.350						
1	218.350	2	154.100	3	52.047	4	3.750	5	2.932
6	249.920	7	1153.200	8	1.000	9	4.610	10	1.151
11	1.000	12	952.300	13	270.760	14	2541.800	15	7179.600
16	295.517	17	1.000	18	278.000	19	2.594	20	1.000
21	655.000	22	1001400.000	23	72000.000	24	52453.000	25	37000.000
26	792700.000	27	31300.000	28	45000.000	29	13100.000	30	52774.000
31	21100.000	32	559800.000	33	220000.000	34	83000.000	35	333000.000
36	.000	37	.000	38	38000.000	39	7500.000	40	530800.000
41	100166.000	42	.000	43	.000	44	4617000.000	45	4618000.000
46	370980.000	47	37100.000	48	2491400.000	49	249100.000	50	5350800.000
51	1070000.000	52	3103900.000	53	310350.000	54	23391720.000	55	2300000.000
56	42000.000	57	4000.000	58	15870.000	59	5350.000	60	.000
61	4777.300	62	11364000.000	63	1136400.000	64	7542.000	65	1.000
66	.000	67	24329000.000	68	1.000	69	24789000.000	70	9240000.000
71	2240.200	72	1.000	73	243000.000	74	7000000.000	75	4670000.000
76	912330.000	77	1.000	78	1.000	79	1.000	80	9123300.000
81	544300.000	82	354900.000	83	5339.712	84	3511.000	85	.000
86	1.000	87	1.000	88	4950000.000	89	9900000.000	90	192.000
91	2379000.000	92	494000.000	93	96800.000	94	.000	95	11897000.000
96	2800000.000	97	1130000.000	98	1.000	99	290000.000	100	.000
101	.000	102	1.000	103	.000	104	.000	105	1.200
106	.000	107	1.000	108	.000	109	.000	110	1.000
111	27000.000	112	90.000	113	100.000	114	.000	115	.000

Table 11.19 CLASS CYCLE LIQUID METAL MHD SYSTEM ACCOUNT LISTING  
PARAMETRIC POINT NO.14

ACCOUNT NO. & NAME	UNIT	AMOUNT	MAT \$/UNIT	INS \$/UNIT	MAT COST \$	INS COST \$
<b>SITE DEVELOPMENT</b>						
1. 1 LAND COST	ACR	142.3	1000.00	.00	132000.00	36396.36
1. 2 CLEARING LAND	ACR	600.7	.00	.00	.00	540000.00
1. 3 GRADING LAND	ACR	142.3	.00	7000.00	.00	540000.00
1. 4 ACCESS RAILROAD	MIL	115000.00	110000.00	.00	575000.00	575000.00
1. 5 LOOP RAILROAD TRACK	MIL	120000.00	70000.00	.00	330000.00	175000.00
1. 6 SIDING R R TRACK	MIL	.00	125000.00	60000.00	.00	.00
1. 7 OTHER SITE COSTS	ACR	.00	.00	.00	337331.29	337331.29
PERCENT TOTAL DIRECT COST IN ACCOUNT 1 =			.674		1444031.28	1654427.64
<b>EXCAVATION &amp; PILING</b>						
2. 1 COMMON EXCAVATION	Y03	135000.0	.00	3.00	.00	407700.00
2. 2 PILING	FT	302400.0	0.50	8.50	2355600.00	3086400.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 2 =			1.255		2355600.00	3493100.00
<b>PLANT ISLAND CONCRETE</b>						
3. 1 PLANT IS. CONCRETE	Y03	10300.0	70.00	80.00	1291000.00	1864000.00
3. 2 SPECIAL STRUCTURES	Y03	27000.0	30.00	150.00	2430000.00	4050000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 3 =			1.981		3711000.00	5914000.00
<b>HEAT REJECTION SYSTEM</b>						
4. 1 COOLING TOWERS	EACH	21.2	.00	.00	3223500.00	1606500.00
4. 2 CIRCULATING H2O SYS	EACH	1.0	.00	.00	1836214.25	240790.34
4. 3 SURFACE CONDENSERS	FT2	313717.3	.00	.00	2434974.12	433102.13
PERCENT TOTAL DIRECT COST IN ACCOUNT 4 =			2.576		7493568.37	4510392.44
<b>STRUCTURAL FEATURES</b>						
5. 1 STAT. STRUCTURAL ST.	TON	2975.0	650.00	175.00	1858750.00	503125.00
5. 2 SILOS & BUNKERS	TPH	385.8	1800.00	750.00	694419.84	285341.64
5. 3 CHIMNEY	FT	400.0	.00	.00	435070.32	552606.38
5. 4 STRUCTURAL FEATURES	EACH	1.0	1034000.00	400000.00	1434000.00	400000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 5 =			1.349		4432240.81	1850073.02
<b>BUILDINGS</b>						
6. 1 STATION BUILDINGS	FT2	734000.0	.10	.10	1174400.00	1174400.00
6. 2 ADMINISTRATION	FT2	10000.0	16.00	13.00	160000.00	140000.00
6. 3 WAREHOUSE & SHOP	FT2	20000.0	12.00	8.00	240000.00	160000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 6 =			.555		1574400.00	1474400.00
<b>FUEL HANDLING &amp; STORAGE</b>						
7. 1 COAL HANDLING SYS	TPH	385.8	.00	.00	5820365.06	2332796.62
7. 2 DOLDRITE HAND. SYS	TPH	57.2	.00	.00	974584.88	515646.30
7. 3 FUEL OIL HAND. SYS	GAL	520000.0	.00	.00	79461.99	67277.91
PERCENT TOTAL DIRECT COST IN ACCOUNT 7 =			0.000		5974410.87	2937716.81
<b>FUEL PROCESSING</b>						
8. 1 COAL DRYER & CRUSHER	TPH	385.8	.00	.00	1370366.59	919577.73
8. 2 CARBONIZERS	TPH	.0	.00	.00	.00	.00
8. 3 CASIFIERS	TPH	.0	.00	.00	.00	.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 8 =			.404		1370366.59	919577.73

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Table 11.19 CLOSURE CYCLE LIQUID METAL PHT SYSTEM ACCOUNT LISTING  
(Continued) PARAMETRIC POINT 40.19

ACCOUNT NO. & NAME	UNIT	AMOUNT	MAT %/UNIT	INS %/UNIT	PAT COST:\$	INS COST:\$
<b>FIRING SYSTEM</b>						
9. 1		.7	.33	.33	.33	.33
PERCENT TOTAL DIRECT COST IN ACCOUNT 9 =		.000	ACCOUNT TOTAL:\$		.00	.00
<b>VAPOR GENERATOR (FIRED)</b>						
10. 1		.3	.33	.00	.33	.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 10 =		.000	ACCOUNT TOTAL:\$		.00	.00
<b>ENERGY CONVERTER</b>						
11. 1	MHD DUCT STRUCT STEEL	KP	191.0	1490000.00	595000.00	1430000.00
11. 2	DUCT INSULATION	KP	132.0	76000.00	20000.00	76000.00
11. 3	RISE	KP	47.0	30000.00	24000.00	50000.00
11. 4	NOZZLE	KP	2.6	33000.00	13000.00	33000.00
11. 5	SEPARATOR/DIFFUSER	KP	2.1	38000.00	15000.00	39000.00
11. 6	LH PIPING/FITTING/INS	KP	190.7	477000.00	171000.00	427000.00
11. 7	IRON BOLTS	KP	313.0	11000.00	26000.00	55100.00
11. 8	AUX EQUIPMENT & SYSTEM		1.0	.00	.00	.00
11. 9	MAGNET DEMAG	KP	3.5	23000.00	13000.00	23000.00
11.10	SUPERCON MAG COOLANT FT3		1.0	456000.00	182000.00	456000.00
11.11	ASSEMBLY SUPPORT STRUCT		1.0	.00	.00	.00
11.12	L H PUMPS	KP	215.0	3070000.00	4000000.00	3970000.00
11.13	L H INVENTORY (SODIUM)KP		215.5	295000.00	30000.00	235000.00
11.14	L H EMERGENCY DUMP	GAL	32373.0	3124000.00	210000.00	2124000.00
11.15	L H PURIFICATION	SPM	5177.0	490000.00	490000.00	490000.00
11.16	GAS COMPRESSOR	KP	264.2	2770000.00	280000.00	2770000.00
11.17	MOTOR DRIVES	MW	527.7	13379000.00	1300000.00	18970000.00
11.18	GAS INVENTORY	KP	19500.0	29000.00	3000.00	29000.00
11.19	GAS PIPING	KP	22.2	11600.00	6000.00	13600.00
11.20	STEAM TURBINE-CLN	MW	30.0	70000.00	70000.00	67000.00
11.21	DRIVE TURBINES	MW	715.0	3410000.00	3400000.00	3410000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 11 =		30.236	ACCOUNT TOTAL:\$		26517400.00	10970000.00
<b>COUPLING HEAT EXCHANGER</b>						
12. 1	COMB HEAT EXCHANGER	KP	5550.0	79440000.00	31730000.00	73440000.00
12. 2	OPEN CYCLE MHD TOPPING		1.0	.00	.00	.00
12. 3	STEAM GENERATOR	KP	4074.2	20360000.00	9400000.00	20360000.00
12. 4	INSULATION		1.0	6000000.00	4000000.00	6000000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 12 =		32.323	ACCOUNT TOTAL:\$		106340000.00	44190000.00
<b>HEAT RECOVERY HEAT EXCH.</b>						
13. 1	AIR PREHEATERS	KP	1925.0	740000.00	800000.00	7840000.00
13. 2	GAS RECUPERATORS	KP	1.0	.00	.00	.00
13. 3	CUMP INTERCOLLER	KP	1.0	.00	.00	.00
13. 4	INSULATION	KP	1.3	170000.00	310000.00	470000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 13 =		3.023	ACCOUNT TOTAL:\$		8310000.00	1130000.00
<b>WATER TREATMENT</b>						
14. 1	DEMINERALIZER	SPM	113.4	2500.00	700.00	298400.00
14. 2	CONDENSATE POLISHING	KWE	74000.0	1.25	.30	932500.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 14 =		.330	ACCOUNT TOTAL:\$		1230399.99	93352.00

Table 11.19 CLOSE CYCLE LIQUID METAL MFG SYSTEM ACCOUNT LISTING  
(Continued) PARAMETRIC POINT NO.14

ACCOUNT NO. & NAME	UNIT	AMOUNT	MAT \$/UNIT	INS \$/UNIT	MAT COST,\$	INS COST,\$
<b>POWER CONDITIONING</b>						
15. 1 INVERTERS		4877.7	43730025.00	8000000.00	45000000.00	3000000.00
15. 2 TRANSFORMERS	KP	3247.4	10757000.00	2200000.00	10557000.00	2200000.00
15. 3 CIRCUIT BREAKERS		172.7	130000.00	20000.00	430000.00	20000.00
15. 4 COOLING SYSTEM		1.0	.00	.00	.00	.00
15. 5 CONDUCTORS		1.2	200000.00	1100000.00	280000.00	1100000.00
15. 6 STD TRANSFORMERS		36666.7	.00	.00	528888.11	10577.36
PERCENT TOTAL DIRECT COST IN ACCOUNT 15 =		15.471	ACCOUNT TOTAL,\$		53715363.00	12330577.25
<b>AUXILIARY MECH EQUIPMENT</b>						
16. 1 BOILER FEED PUMP & DR	KWE	723512.5	1.67	.10	1208285.87	72351.25
16. 2 OTHER PUMPS	KWE	293955.8	.82	.12	256041.36	34914.81
16. 3 MISC SERVICE SYS	KWE	969888.9	1.17	.73	1134731.39	767994.80
16. 4 AUXILIARY BOILER	PP4	293000.0	4.30	.80	1130000.00	224000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 16 =		1.022	ACCOUNT TOTAL,\$		3719039.22	1035260.84
<b>PIPE &amp; FITTINGS</b>						
17. 1 CONVENTIONAL PIPING	TON	3300.3	3000.00	1200.00	11400000.00	6840000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 17 =		2.117	ACCOUNT TOTAL,\$		11400000.00	6840000.00
<b>AUXILIARY ELEC EQUIPMENT</b>						
18. 1 MISC NOTES, ETC		230850.8	1.47	.17	477339.49	43462.55
18. 2 SWITCHGEAR & MCC PAN	KWE	327842.4	1.98	.45	754487.59	174574.06
18. 3 CONDUIT, CABLES, TRAYS	FT	4325000.0	1.32	1.35	5734739.94	590393.94
18. 4 ISOLATED PHASE BUS	FT	500.0	510.00	450.00	255000.00	225000.00
18. 5 LIGHTING & COMMUN	KWE	383855.3	.35	.43	339449.56	417338.04
PERCENT TOTAL DIRECT COST IN ACCOUNT 18 =		7.090	ACCOUNT TOTAL,\$		7558278.58	6836474.52
<b>CONTROL &amp; INSTRUMENTATION</b>						
19. 1 COMPUTER	EACH	1.0	60000.00	10000.00	50000.00	10000.00
19. 2 OTHER CONTROLS	EACH	1.0	100000.00	50000.00	100000.00	50000.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 19 =		.512	ACCOUNT TOTAL,\$		177000.00	65000.00
<b>PROCESS WASTE SYSTEMS</b>						
20. 1 BOTTOM ASH	TPH	29.6	300095.74	772748.84	300095.34	772748.84
20. 2 DRY ASH	TPH	7.4	18430.12	134122.53	535495.12	134122.53
20. 3 WET SLURRY	TPH	57.5	10606.62	402871.66	181080.62	402871.66
20. 4 ONSITE DISPOSAL	ACRE	707.5	8352.40	9473.47	1956337.72	2914479.34
20. 5 SEED TREATMENT	EACH	.0	.00	.00	.00	.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 20 =		2.452	ACCOUNT TOTAL,\$		7134503.75	4224022.81
<b>STACK GAS CLEANING</b>						
21. 1 PRECIPITATOR	EACH	.0	7191279.69	4667831.75	.00	.00
21. 2 SCRUBBER	KWE	100000.0	22.18	13.13	2200000.00	10130781.12
21. 3 MISC STEEL & DUCTS		.0	.00	.00	.00	.00
PERCENT TOTAL DIRECT COST IN ACCOUNT 21 =		3.329	ACCOUNT TOTAL,\$		2200000.00	10130781.12
<b>TOTAL DIRECT COSTS,\$</b>					<b>34453953.00</b>	<b>121006151.00</b>

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Table 11.20 CLOSE CYCLE LIQUID METAL MHD SYSTEM COST OF ELECTRICITY, MILLS/KW.HR  
PARAMETRIC POINT NO.19

ACCOUNT	RATE, PERCENT	6.00	LABOR RATE, 1/HR F.50	10.00	15.00	21.50
TOTAL DIRECT COSTS,\$	.0	417163732.	421702923.	455675835.	515904804.	530105698.
INDIRECT COST,\$	51.0	34931863.	42406848.	61713136.	87329909.	125172869.
PROF & OWNER COSTS,\$	3.0	31253793.	25335233.	37254066.	41272394.	47238535.
CONTINGENCY COST,\$	11.0	45448010.	48587320.	51224341.	5674528.	64911735.
SUB TOTAL,\$	.0	525595795.	575113415.	615867368.	701256624.	827339808.
ESCALATION COST,\$	0.5	18438070E.	201368140.	215637582.	245535468.	289702788.
INTREST DURING CONST,\$	10.0	234976342.	256625923.	274811072.	312913292.	359200644.
TOTAL CAPITALIZATION,\$	.0	245854432.	107310740.	1106316016.	1259705376.	1486303232.
COST OF ELEC-CAPITAL	18.0	31.43313	36.40332	35.33763	41.94513	49.49036
COST OF ELEC-FUEL	.0	7.45757	7.45757	7.45757	7.45757	7.45757
COST OF ELEC-OP & MAIN	.0	.90212	.90212	.90212	.90212	.90212
TOTAL COST OF ELEC	.0	30.85772	42.75870	45.19737	50.30487	57.85004

ACCOUNT	RATE, PERCENT	-5.00	CONTINGENCY, PERCENT .00	11.00	5.00	20.00
TOTAL DIRECT COSTS,\$	.0	465675836.	415675836.	465675836.	421675836.	465675836.
INDIRECT COST,\$	51.0	61713136.	61713136.	61713136.	61713136.	61713136.
PROF & OWNER COSTS,\$	3.0	37254066.	37254066.	37254066.	37254066.	37254066.
CONTINGENCY COST,\$	20.0	-23283791.	.0	51224341.	23283791.	93135166.
SUB TOTAL,\$	.0	541359248.	564643032.	615867368.	56792810.	657778192.
ESCALATION COST,\$	5.5	189549575.	197702079.	215637532.	205954578.	230312086.
INTREST DURING CONST,\$	10.0	241564214.	251953856.	274811072.	252343498.	253512436.
TOTAL CAPITALIZATION,\$	.0	372473032.	1014298951.	1105316016.	1056124983.	1181602704.
COST OF ELEC-CAPITAL	18.0	32.38103	33.77374	36.83763	35.16644	39.34455
COST OF ELEC-FUEL	.0	7.45757	7.45757	7.45757	7.45757	7.45757
COST OF ELEC-OP & MAIN	.0	.90212	.90212	.90212	.90212	.90212
TOTAL COST OF ELEC	.0	40.74072	42.13343	45.19737	43.52613	47.70424

ACCOUNT	RATE, PERCENT	5.00	ESCALATION RATE, PERCENT 6.50	8.00	10.00	.00
TOTAL DIRECT COSTS,\$	.0	455675835.	455675835.	455675835.	455675835.	455675835.
INDIRECT COST,\$	51.0	61713136.	61713136.	61713136.	61713136.	61713136.
PROF & OWNER COSTS,\$	3.0	37254066.	37254066.	37254066.	37254066.	37254066.
CONTINGENCY COST,\$	11.0	51224341.	51224341.	51224341.	51224341.	51224341.
SUB TOTAL,\$	.0	515967368.	615867368.	615867368.	615867368.	615867368.
ESCALATION COST,\$	0.0	160618574.	215637582.	274118968.	357852072.	0.
INTREST DURING CONST,\$	10.0	259594764.	274811072.	230809996.	313417584.	214115582.
TOTAL CAPITALIZATION,\$	.0	1036080696.	117316016.	1180796320.	1297087024.	829983748.
COST OF ELEC-CAPITAL	18.0	34.49922	35.83759	39.31770	42.85693	27.53646
COST OF ELEC-FUEL	.0	7.45757	7.45757	7.45757	7.45757	7.45757
COST OF ELEC-OP & MAIN	.0	.90212	.90212	.90212	.90212	.90212
TOTAL COST OF ELEC	.0	42.85871	45.19737	47.67739	51.21662	35.99614

ACCOUNT	RATE, PERCENT	6.00	INT DURING CONST, PERCENT 3.00	10.00	12.50	15.00
TOTAL DIRECT COSTS,\$	.0	465675836.	465675836.	465675836.	465675836.	465675836.
INDIRECT COST,\$	51.0	61713136.	61713136.	61713136.	61713136.	61713136.
PROF & OWNER COSTS,\$	3.0	37254066.	37254066.	37254066.	37254066.	37254066.
CONTINGENCY COST,\$	11.0	51224341.	51224341.	51224341.	51224341.	51224341.
SUB TOTAL,\$	.0	615867368.	615867368.	615867368.	615867368.	615867368.
ESCALATION COST,\$	5.5	215637582.	215637582.	215637582.	215637582.	215637582.
INTREST DURING CONST,\$	15.0	156145254.	219226678.	274811072.	355513720.	441616184.
TOTAL CAPITALIZATION,\$	.0	387550792.	1745431616.	1105316016.	117013555.	1273121120.
COST OF ELEC-CAPITAL	18.0	32.83642	34.81738	36.83769	39.52489	42.39190
COST OF ELEC-FUEL	.0	7.45757	7.45757	7.45757	7.45757	7.45757
COST OF ELEC-OP & MAIN	.0	.90212	.90212	.90212	.90212	.90212
TOTAL COST OF ELEC	.0	41.24610	43.17007	45.19737	47.89458	50.75159

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Table 11.20 CLOSE CYCLE LIQUID METAL PNC SYSTEM COST OF ELECTRICITY, MILLS/KW.H  
(Continued) PARAMETRIC POINT 10.14

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ACCOUNT	RATE, PERCENT	FIXED CHARGE RATE, PCT				
		10.00	14.40	18.00	21.60	25.00
TOTAL DIRECT COSTS,\$	0.0	455675933.	455675933.	455675933.	455675933.	455675933.
INDIRECT COSTS,\$	51.0	61713136.	61713136.	61713136.	61713136.	61713136.
PROP & OWNER COSTS,\$	4.0	37254066.	37254066.	37254066.	37254066.	37254066.
CONTINGENCY COSTS,\$	11.0	51224341.	51224341.	51224341.	51224341.	51224341.
SUB TOTAL,\$	0.0	515867369.	515867369.	515867369.	515867369.	515867369.
ESCALATION COST,\$	0.0	215637582.	215637582.	215637582.	215637582.	215637582.
INTEREST DURING CONST. \$	15.0	274911072.	274911072.	274911072.	274911072.	274911072.
TOTAL CAPITALIZATION,\$	0.0	1106316016.	1106316016.	1106316016.	1106316016.	1106316016.
COST OF ELEC-CAPITAL	15.0	36.83769	36.83769	36.83769	36.83769	36.83769
COST OF ELEC-FUEL	0.0	7.45757	7.45757	7.45757	7.45757	7.45757
COST OF ELEC-OP & MAINT	0.0	.90212	.90212	.90212	.90212	.90212
TOTAL COST OF ELEC	0.0	20.82507	37.82889	45.18737	52.56491	59.57314

ACCOUNT	RATE, PERCENT	FUEL COST, %/10**E BTU				
		1.50	1.55	2.50	1.00	1.07
TOTAL DIRECT COSTS,\$	0.0	455675933.	455675933.	455675933.	455675933.	455675933.
INDIRECT COSTS,\$	51.0	61713136.	61713136.	61713136.	61713136.	61713136.
PROP & OWNER COSTS,\$	4.0	37254066.	37254066.	37254066.	37254066.	37254066.
CONTINGENCY COSTS,\$	11.0	51224341.	51224341.	51224341.	51224341.	51224341.
SUB TOTAL,\$	0.0	515867369.	515867369.	515867369.	515867369.	515867369.
ESCALATION COST,\$	0.0	215637582.	215637582.	215637582.	215637582.	215637582.
INTEREST DURING CONST. \$	10.0	274911072.	274911072.	274911072.	274911072.	274911072.
TOTAL CAPITALIZATION,\$	0.0	1106316016.	1106316016.	1106316016.	1106316016.	1106316016.
COST OF ELEC-CAPITAL	15.0	36.83769	36.83769	36.83769	36.83769	36.83769
COST OF ELEC-FUEL	0.0	4.34631	7.45757	13.16042	21.93403	8.34909
COST OF ELEC-OP & MAINT	0.0	.90212	.90212	.90212	.90212	.90212
TOTAL COST OF ELEC	0.0	42.12651	45.19737	55.90022	59.67384	48.68889

ACCOUNT	RATE, PERCENT	CAPACITY FACTOR, PERCENT				
		45.00	50.00	55.00	60.00	80.00
TOTAL DIRECT COSTS,\$	0.0	455675933.	455675933.	455675933.	455675933.	455675933.
INDIRECT COSTS,\$	51.0	61713136.	61713136.	61713136.	61713136.	61713136.
PROP & OWNER COSTS,\$	4.0	37254066.	37254066.	37254066.	37254066.	37254066.
CONTINGENCY COSTS,\$	11.0	51224341.	51224341.	51224341.	51224341.	51224341.
SUB TOTAL,\$	0.0	515867369.	515867369.	515867369.	515867369.	515867369.
ESCALATION COST,\$	0.0	215637582.	215637582.	215637582.	215637582.	215637582.
INTEREST DURING CONST. \$	15.0	274911072.	274911072.	274911072.	274911072.	274911072.
TOTAL CAPITALIZATION,\$	0.0	1106316016.	1106316016.	1106316016.	1106316016.	1106316016.
COST OF ELEC-CAPITAL	15.0	36.83769	36.83769	36.83769	36.83769	36.83769
COST OF ELEC-FUEL	0.0	7.45757	7.45757	7.45757	7.45757	7.45757
COST OF ELEC-OP & MAINT	0.0	.90212	.90212	.90212	.90212	.90212
TOTAL COST OF ELEC	0.0	20.82507	61.56969	56.24868	45.16737	38.29031

Table 11.21 CLOSE CYCLE LIQUID METAL MHD SYSTEM

ACCOUNT NO	AUX POWER, MWE	PERC PLANT POW	OPERATION COST	MAINTENANCE COST
4	14.57355	1.52779	89.31764	20.53764
7	4.27377	.44545	307.53159	.00000
9	.57153	.05283	.00000	.00000
14	.00000	.00000	12.91023	.00000
18	5.19000	.54532	.00000	.00000
20	9.10720	.94875	.00000	.00000
21	17.00000	1.75000	.00000	.00000
TOTALS	50.61606	5.33146	556.44991	20.53764

CLOSE CYCLE LIQUID METAL MHD SYSTEM		BASE CASE INPUT	
NOMINAL POWER, MWE	1000.0000	NET POWER, MWE	949.3839
NOM HEAT RATE, BTU/KW-HR	8329.5231	NET HEAT RATE, BTU/KW-HR	3773.6134
OFF DESIGN HEAT RATE	.0000		
CONDENSER DESIGN PRESSURE, IN HG A	3.5000	NUMBER OF SHELLS	2.0000
NUMBER OF TUBES/SHELL	17000.7139	TUBE LENGTH, FT	59.5067
G, BTU/HR-FT <sup>2</sup> -F	609.9535	TERMINAL TEMP DIFF, F	1.0000
HEAT REJECTION DESIGN TEMP, F	77.0000	APPROACH, F	15.6713
RANGE, F	23.0000	OFF DESIGN TEMP, F	51.4000
OFF DESIGN PRES, IN HG A	.0000	LP TURBINE BLADE LEN, IN	28.5000

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1	1000.000	2	.000	3	.410	4	.000	5	8.000
6	30.000	7	3.500	8	1603.000	9	2.000	10	2.000
11	1.000	12	713.000	13	1.000	14	.000	15	.000
16	2.000	17	182.000	18	3.000	19	5.000	20	2.500
21	.000	22	19300.000	23	27000.000	24	2975.000	25	430.000
26	7340000.000	27	16000.000	28	20000.000	29	525000.000	30	.300
31	1.000	32	3300.000	33	1.000	34	.300	35	.800
36	4390000.000	37	500.000	38	1.000	39	1.000	40	1434000.000
41	405000.000	42	550000.000	43	15000.000	44	1050000.000	45	641000.000
46	6.000	47	.000	48	.000	49	2.000	50	2.000
51	1.000	52	5.350						
56	181.000	57	132.000	58	40.000	59	2.700	60	2.146
61	190.741	62	913.000	63	1.000	64	3.499	65	.990
66	1.000	67	918.000	68	215.500	69	32373.000	70	6167.000
71	254.271	72	327.000	73	185000.000	74	22.204	75	30.000
76	716.000	77	1490000.000	78	590000.000	79	72000.000	80	20000.000
81	500000.000	82	240000.000	83	33000.000	84	13000.000	85	39000.000
86	16000.000	87	4270000.000	88	1710000.000	89	60000.000	90	26000.000
91	.000	92	.000	93	29800.000	94	12000.000	95	456000.000
96	192000.000	97	.000	98	.000	99	3970000.000	100	4000000.000
101	295000.000	102	30000.000	103	2120000.000	104	210000.000	105	4880000.000
106	490000.000	107	2770000.000	108	780000.000	109	13970000.000	110	1900000.000
111	23000.000	112	3000.000	113	13600.000	114	5000.000	115	670000.000
116	70000.000	117	9410000.000	118	940000.000	119	6560.000	120	1.000
121	4074.200	122	1.000	123	79440000.000	124	31730000.000	125	.000
126	.000	127	20000000.000	128	8400000.000	129	6000000.000	130	4000000.000
131	1925.000	132	1.000	133	1.000	134	1.000	135	7840000.000
136	800000.000	137	.000	138	.000	139	.000	140	.000
141	470000.000	142	310000.000	143	4870.000	144	3293.400	145	172.000
146	1.000	147	1.000	148	45000000.000	149	9000000.000	150	10957000.000
151	2200000.000	152	430000.000	153	20000.000	154	.000	155	.000
156	2800000.000	157	1100000.000	158	1.000	159	280000.000	160	1.200
161	.000	162	1.000	163	.000	164	.000	165	1.000
166	27000.000	167	50.000	168	150.000	169	.000	170	.000

Cost of L M-MHD Systems & Components , % of Total

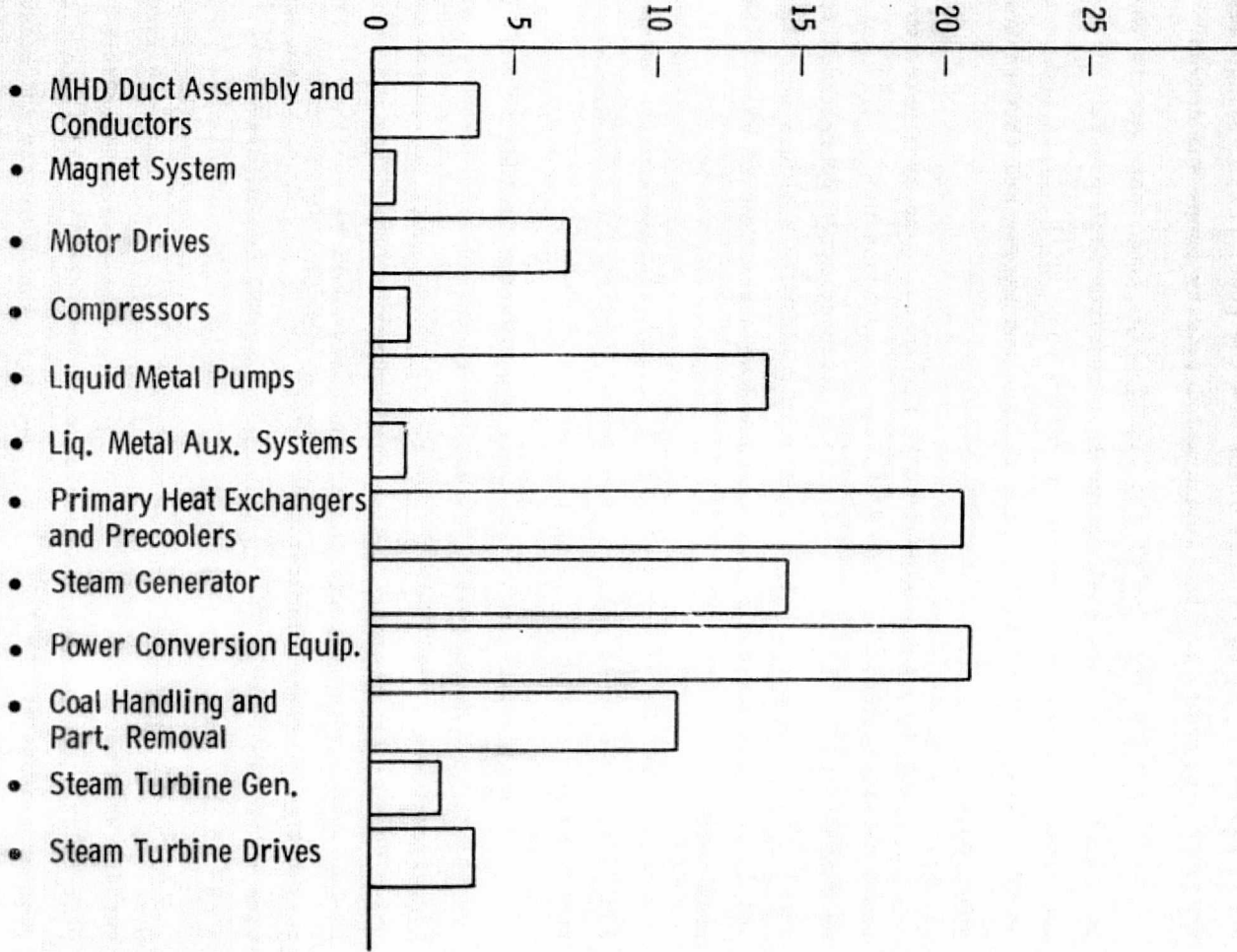


Fig. 11.11 - Comparison of LM-MHD major component costs - base case



The costs associated with the liquid-metal subsystems such as the purification, emergency dump, and storage are also small compared to those of other systems. The major cost items are the power conditioning, heat transfer equipment, and the liquid-metal primary pumps and drives.

A large fraction of the liquid-metal pump costs are in the so-called free surface seal design that requires high-pressure tankage and cross-over piping. Developing sodium pumps with direct sodium seals would eliminate the need for this tankage and piping and thus reduce pump costs.

The need for a liquid-metal mechanical pump as a prime mover could be eliminated if a high-performance two-phase nozzle diffuser could be demonstrated. The two-phase nozzle diffuser would be required to operate for long periods efficiently at relatively high gas void fractions and large flows but could be constructed at a fraction of the mechanical pump costs.

Both the direct sodium seal mechanical pump and the two-phase nozzle diffuser would require significant engineering development and, therefore, are not considered as near term.

High power conversion costs are incurred because of the large low-voltage MHD gross power that must be generated in order to produce significant net power. All the low-voltage power must be treated by the power conversion equipment. With the homogenous flow, two-component LM-MHD system, reduced power conversion costs could be realized through more cost-effective power conversion equipment and if highly efficient MHD ducts were to be developed. The application of ac-MHD generators was beyond the scope of this work.

The high heat transfer equipment costs arise from the large surface area needed to transfer heat in gas-to-gas type exchangers, particularly the primary heat exchanger. This problem was further compounded because of the large flows needed in the MHD cycle and because argon gas is not a particularly effective heat transfer medium. Thus, a trade-off must be made between the thermodynamics of the cycle and heat

transfer characteristics of the MHD working fluids. Development of improved high-temperature metals technology should also lead to reduced heat exchanger costs.

## 11.6 Analysis of Overall Cost of Electricity

### 11.6.1 Cost Accounts

Included in the costing of each parametric point were some 21 accounts plus summations and additions leading to the cost of electricity for various capacity factors, fixed charges, labor rates, interest during construction, escalation rate, and fuel costs. A breakdown of these accounts was provided as computer printout for each case. Tables 11.13, 11.16, and 11.19 show these printouts for the base case (Point 16), Points 8 and 14, respectively. Summary sheets for all costed points are shown in Tables 11.22, 11.23, and 11.24. The LM-MHD natural resource requirements for each costed study point are given in Table 11.25. These resource tables give a quantitative breakdown of the water and land usage and ultimate fuel requirements. The major environmental intrusions are associated with the coal ash handling and disposal and with the stack-gas desulfurization, as itemized in Accounts 20 and 21 of Tables 11.13, 11.16, and 11.19. Cleanup of these power plants is not expected to present formidable problems or costs that exceed current coal-fired steam plant technology.

### 11.6.2 Comparison of the Cost of Electricity for LM-MHD Steam Binary Power Plant

The cost of electricity for each costed parametric point are given in Table 11.22. A bar chart comparison is shown in Figure 11.12 of these costs for five representative binary plant study cases. Capital, fuel, operating and maintenance, and total costs are shown. For the cases shown, the costs vary from a low of about 9.44 mills/MJ (34 mills/kWh) to slightly over 12.78 mills/MJ (46 mills/kWh). The higher cost cases correspond to the 1088°K (1500°F) duct inlet temperature. The lowest cost was Point 12, a parametric variation of the base case that utilized the two-phase nozzle diffuser as the prime mover for the liquid metal. The

Table 11.22 CLOSE CYCLE LIQUID METAL MHD SYSTEM SUMMARY PLANT RESULTS

PARAMETRIC POINT	1	2	3	4	5	6	7	8
THERMODYNAMIC EFF	.333	.333	.333	.333	.333	.333	.333	.333
POWER PLANT EFF	.600	.600	.600	.294	.600	.371	.600	.358
OVERALL ENERGY EFF	.333	.333	.333	.334	.333	.371	.333	.358
CAP COST MILLION \$	.000	.000	.000	1315	.000	1722	.000	1112
CAPITAL COST \$/KWE	.000	.000	.000	322	.000	17	.000	177
COE CAPITAL	.000	.000	.000	44	.000	57	.000	37
COE FUEL	.333	.333	.333	9	.333	7	.333	8
COE OP & MAINT	.000	.000	.000	.001	.000	.001	.000	.001
COST OF ELECTRIC	.033	.033	.033	54	.033	55	.033	46
EST TIME OF CONST	.000	.000	.000	8.000	.000	8.000	.000	8.000

PARAMETRIC POINT	9	10	11	12	13	14	15	16
THERMODYNAMIC EFF	.333	.333	.333	.333	.333	.333	.333	.333
POWER PLANT EFF	.371	.600	.342	.360	.305	.339	.600	.337
OVERALL ENERGY EFF	.371	.600	.342	.360	.305	.339	.600	.337
CAP COST MILLION \$	809	194	854	464	751	122	2039	277
CAPITAL COST \$/KWE	351	719	373	793	193	133	739	155
COE CAPITAL	25	925	29	472	4	380	67	644
COE FUEL	7	825	3	470	3	512	7	458
COE OP & MAINT	.501	.000	.332	.315	.622	.302	.000	.337
COST OF ELECTRIC	35	555	37	875	73	839	45	197
EST TIME OF CONST	8.000	.000	8.000	8.000	8.000	8.000	.000	6.000

PARAMETRIC POINT	17	18	19	20	21	22	23	24
THERMODYNAMIC EFF	.333	.333	.333	.333	.333	.333	.333	.333
POWER PLANT EFF	.600	.347	.600	.600	.600	.600	.600	.600
OVERALL ENERGY EFF	.600	.347	.600	.600	.600	.600	.600	.600
CAP COST MILLION \$	.000	118	.000	.000	.000	.000	.000	.000
CAPITAL COST \$/KWE	.000	151	.000	.000	.000	.000	.000	.000
COE CAPITAL	.000	36	.000	.000	.000	.000	.000	.000
COE FUEL	.333	3	.333	.333	.333	.333	.333	.333
COE OP & MAINT	.000	1	.000	.000	.000	.000	.000	.000
COST OF ELECTRIC	.033	45	.033	.033	.033	.033	.033	.033
EST TIME OF CONST	.000	8.000	.000	.000	.000	.000	.000	.000

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Table 11.23

## CLOSE CYCLE LIQUID METAL MFC SYSTEM SUMMARY PLANT RESULTS

PARAMETRIC POINT	1	2	3	4	5	6	7	P
TOTAL CAPITAL COST	\$.77	\$.30	\$.35	1313.21	\$.00	1723.43	\$.00	1118.40
P LIQ MET DUCT ASSEMBLY	\$.007	\$.000	\$.000	1.656	\$.000	5.714	\$.000	8.383
L SUPERCONDUCTING MAGNET ASSY	\$.000	\$.000	\$.000	6.643	\$.000	3.954	\$.000	1.398
A LIQ MET SUBSYSTEMS	\$.000	\$.000	\$.000	20.366	\$.000	54.820	\$.000	54.369
N LIQ MET PUMP MOTOR DRIVE	\$.000	\$.000	\$.000	18.103	\$.000	21.353	\$.000	23.352
T COMPRESSOR TURBINE DRIVES	\$.000	\$.000	\$.000	6.000	\$.000	10.433	\$.000	11.364
STEAM TURBINE GENERATOR	\$.000	\$.000	\$.000	6.000	\$.000	1.230	\$.000	6.000
COUPLING HEAT EXCHANGERS	\$.000	\$.000	\$.000	265.360	\$.000	285.592	\$.000	133.479
POWER CONDITION EQUIPMENT	\$.000	\$.000	\$.000	33.018	\$.000	53.919	\$.000	55.210
R TOT MAJOR COMPONENT COST	\$.000	\$.000	\$.000	347.171	\$.000	450.580	\$.000	297.615
S TOT MAJOR COMPONENT COST, \$/KWE	\$.000	\$.000	\$.000	365.549	\$.000	473.771	\$.000	313.231
U BALANCE OF PLANT COST	\$.000	\$.000	\$.000	63.355	\$.000	88.910	\$.000	91.675
L SITE LABOR	\$.000	\$.000	\$.000	133.377	\$.000	201.216	\$.000	102.015
T TOTAL DIRECT COST	\$.000	\$.000	\$.000	593.821	\$.000	763.852	\$.000	566.921
INDIRECT COSTS	\$.000	\$.000	\$.000	39.277	\$.000	122.620	\$.000	52.028
PROF & OWNER COSTS	\$.000	\$.000	\$.000	47.511	\$.000	61.106	\$.000	40.554
CONTINGENCY COST	\$.000	\$.000	\$.000	55.327	\$.000	84.024	\$.000	55.761
ESCALATION COST	\$.000	\$.000	\$.000	271.354	\$.000	254.200	\$.000	229.432
INT DURING CONSTRUCTION	\$.000	\$.000	\$.000	343.817	\$.000	451.337	\$.000	292.350
A TOTAL CAPITALIZATION	\$.000	\$.000	\$.000	1352.166	\$.000	1517.202	\$.000	1177.085
K COST OF ELEC-CAPITAL, MILLS/KWE	\$.000	\$.000	\$.000	44.010	\$.000	57.446	\$.000	37.210
D COST OF ELEC-FUEL, MILLS/KWE	\$.000	\$.000	\$.000	5.893	\$.000	7.810	\$.000	8.098
O COST OF ELEC-OP&MAINT, MILLS/KWE	\$.000	\$.000	\$.000	9.901	\$.000	9.901	\$.000	9.921
W TOTAL COST OF ELEC, MILLS/KWE	\$.000	\$.000	\$.000	54.793	\$.000	66.157	\$.000	46.229
N COE 0.5 CAP. FACTOR, MILLS/KWE	\$.000	\$.000	\$.000	57.346	\$.000	63.393	\$.000	57.393
COE 0.8 CAP. FACTOR, MILLS/KWE	\$.000	\$.000	\$.000	46.542	\$.000	55.386	\$.000	39.253
COE 1.2XCAP. COST, MILLS/KWE	\$.000	\$.000	\$.000	53.535	\$.000	77.645	\$.000	53.671
COE 1.2XFUEL COST, MILLS/KWE	\$.000	\$.000	\$.000	56.770	\$.000	67.719	\$.000	47.849
COE (CONTINGENCY=3), MILLS/KWE	\$.000	\$.000	\$.000	51.034	\$.000	61.335	\$.000	43.053
COE (ESCALATION=0), MILLS/KWE	\$.000	\$.000	\$.000	43.801	\$.000	51.206	\$.000	36.935

PARAMETRIC POINT	9	10	11	12	13	14	15	15
TOTAL CAPITAL COST	333.19	\$.00	354.48	751.12	2339.23	1106.32	\$.00	394.47
P LIQ MET DUCT ASSEMBLY	2.671	\$.000	2.012	3.155	30.092	6.511	\$.000	3.444
L SUPERCONDUCTING MAGNET ASSY	3.354	\$.000	3.923	6.895	5.273	1.145	\$.000	1.049
A LIQ MET SUBSYSTEMS	35.172	\$.000	29.211	5.980	117.660	46.959	\$.000	41.032
N LIQ MET PUMP MOTOR DRIVE	14.059	\$.000	6.300	3.132	52.135	13.970	\$.000	1.740
T COMPRESSOR TURBINE DRIVES	7.522	\$.000	6.926	11.090	6.000	5.410	\$.000	9.170
STEAM TURBINE GENERATOR	6.229	\$.000	6.210	6.000	6.000	5.670	\$.000	5.690
COUPLING HEAT EXCHANGERS	55.567	\$.000	71.038	61.556	336.717	114.650	\$.000	73.838
POWER CONDITION EQUIPMENT	47.245	\$.000	43.143	44.905	155.333	54.715	\$.000	49.992
R TOT MAJOR COMPONENT COST	168.372	\$.000	167.314	134.716	692.016	252.072	\$.000	165.955
S TOT MAJOR COMPONENT COST, \$/KWE	177.213	\$.000	170.354	141.723	733.471	271.331	\$.000	196.022
U BALANCE OF PLANT COST	87.287	\$.000	91.755	90.795	76.402	91.215	\$.000	92.017
L SITE LABOR	93.742	\$.000	107.253	35.393	132.387	127.453	\$.000	107.134
T TOTAL DIRECT COST	358.254	\$.000	375.334	228.512	944.261	490.503	\$.000	395.172
INDIRECT COSTS	47.312	\$.000	54.730	43.357	67.517	55.093	\$.000	54.638
PROF & OWNER COSTS	25.660	\$.000	30.031	26.281	75.541	50.240	\$.000	31.618
CONTINGENCY COST	39.413	\$.000	81.932	35.135	103.863	33.955	\$.000	43.453
ESCALATION COST	166.012	\$.000	175.561	154.020	417.078	227.134	\$.000	183.784
INT DURING CONSTRUCTION	211.557	\$.000	223.737	135.235	531.523	283.453	\$.000	234.217
A TOTAL CAPITALIZATION	851.714	\$.000	960.705	790.190	2139.795	1165.223	\$.000	942.894
K COST OF ELEC-CAPITAL, MILLS/KWE	26.925	\$.000	23.473	24.330	57.644	36.333	\$.000	29.377
D COST OF ELEC-FUEL, MILLS/KWE	7.923	\$.000	6.470	8.052	2.512	7.458	\$.000	8.609
O COST OF ELEC-OP&MAINT, MILLS/KWE	9.921	\$.000	9.901	9.901	9.901	9.901	\$.000	9.921
W TOTAL COST OF ELEC, MILLS/KWE	31.625	\$.000	37.225	33.347	78.038	45.197	\$.000	39.354
N COE 0.5 CAP. FACTOR, MILLS/KWE	43.732	\$.000	45.517	41.441	93.371	56.243	\$.000	48.235
COE 0.8 CAP. FACTOR, MILLS/KWE	30.606	\$.000	32.536	29.263	65.355	38.250	\$.000	33.765
COE 1.2XCAP. COST, MILLS/KWE	41.043	\$.000	43.570	33.943	91.557	52.555	\$.000	45.315
COE 1.2XFUEL COST, MILLS/KWE	37.221	\$.000	39.569	35.557	70.940	46.629	\$.000	41.075
COE (CONTINGENCY=3), MILLS/KWE	33.417	\$.000	35.533	31.835	72.133	42.133	\$.000	36.895
COE (ESCALATION=0), MILLS/KWE	26.923	\$.000	30.713	27.706	61.142	35.956	\$.000	31.908

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Table 11.23  
(Continued)

CLOSE CYCLE LIQUID METAL MHD SYSTEM SUMMARY PLANT RESULTS

PARAMETRIC POINT	17	18	19	20	21	22	23	24
TOTAL CAPITAL COST	.00	1119.95	.00	.00	.00	.00	.00	.00
LIG MET DUCT ASSEMBLY	.000	3.444	.000	.000	.000	.000	.000	.000
SUPERCONDUCTING MAGNET ASSY	.000	1.049	.000	.000	.000	.000	.000	.000
LIG MET SUBSYSTEMS	.000	41.022	.000	.000	.000	.000	.000	.000
LIG MET PUMP MOTOR DRIVE	.000	1.740	.000	.000	.000	.000	.000	.000
COMPRESSOR TURBINE DRIVES	.000	9.176	.000	.000	.000	.000	.000	.000
STEAM TURBINE GENERATOR	.000	5.599	.000	.000	.000	.000	.000	.000
COUPLING HEAT EXCHANGERS	.000	1.347	.000	.000	.000	.000	.000	.000
POWER CONDITION EQUIPMENT	.000	49.992	.000	.000	.000	.000	.000	.000
TOT MAJOR COMPONENT COST	.000	75.555	.000	.000	.000	.000	.000	.000
TOT MAJOR COMPONENT COST	.000	272.416	.000	.000	.000	.000	.000	.000
BALANCE OF PLANT COST	.000	70.961	.000	.000	.000	.000	.000	.000
SITE LABOR	.000	143.222	.000	.000	.000	.000	.000	.000
TOTAL DIRECT COST	.000	474.677	.000	.000	.000	.000	.000	.000
INDIRECT COSTS	.000	75.193	.000	.000	.000	.000	.000	.000
PROF & OWNER COSTS	.000	17.958	.000	.000	.000	.000	.000	.000
CONTINGENCY COST	.000	32.275	.000	.000	.000	.000	.000	.000
ESCALATION COST	.000	27.402	.000	.000	.000	.000	.000	.000
INT DURING CONSTRUCTION	.000	20.971	.000	.000	.000	.000	.000	.000
TOTAL CAPITALIZATION	.000	117.111	.000	.000	.000	.000	.000	.000
COST OF ELEC-CAPITAL	.000	73.393	.000	.000	.000	.000	.000	.000
COST OF ELEC-FUEL	.000	3.361	.000	.000	.000	.000	.000	.000
COST OF ELEC-OP&MAINT	.000	1.842	.000	.000	.000	.000	.000	.000
TOTAL COST OF ELEC	.000	46.596	.000	.000	.000	.000	.000	.000
COE 0.5 CAP. FACTOR	.000	57.514	.000	.000	.000	.000	.000	.000
COE 0.8 CAP. FACTOR	.000	59.772	.000	.000	.000	.000	.000	.000
COE 1.2XCAP. COST	.000	51.875	.000	.000	.000	.000	.000	.000
COE 1.2XFUEL COST	.000	48.268	.000	.000	.000	.000	.000	.000
COE (CONTINGENCY=1)	.000	43.631	.000	.000	.000	.000	.000	.000
COE (ESCALATION=0)	.000	57.506	.000	.000	.000	.000	.000	.000

Table 11.24 CLOSE CYCLE LIQUID METAL MHD SYSTEM NATURAL RESOURCE REQUIREMENTS

PARAMETRIC POINT	1	2	3	4	5	6	7	8
COAL, LB/KW-HR	.0000	.3800	.1700	1.37773	.0000	.35171	.0000	.38315
SORBANT OR SEED, LB/KW-HR	.0000	.0000	.0000	.16058	.0000	.12674	.0000	.13153
TOTAL WATER, GAL/KW-HR	.000	.000	.000	.055	.000	1.323	.000	1.038
COOLING WATER	.000	.000	.000	.000	.000	.966	.000	1.038
GASIFIER PROCESS H2O	.0000	.0000	.0000	.17000	.0000	.0000	.0000	.0000
CONDENSATE MAKE UP	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000
WASTE HANDLING SLURRY	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000
SCRUBBER WASTE WATER	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000
NOX SUPPRESSION	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000
TOTAL LAND ACRES/100MWE	.00	.00	.00	130.80	.00	85.54	.00	88.78
MAIN PLANT	.00	.00	.00	29.53	.00	19.13	.00	19.15
DISPOSAL LAND	.00	.00	.00	42.94	.00	33.51	.00	35.18
LAND FOR ACCESS RR	.00	.00	.00	58.33	.00	32.90	.00	34.44
PARAMETRIC POINT	9	10	11	12	13	14	15	16
COAL, LB/KW-HR	.85331	.0000	.0000	.87810	1.03731	.81372	.0000	.93689
SORBANT OR SEED, LB/KW-HR	.12856	.0000	.0000	.13939	.15456	.12109	.0000	.13918
TOTAL WATER, GAL/KW-HR	1.828	.000	.000	1.062	.000	1.222	.000	1.022
COOLING WATER	.588	.000	.000	.000	.000	1.175	.000	.957
GASIFIER PROCESS H2O	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000
CONDENSATE MAKE UP	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000
WASTE HANDLING SLURRY	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000
SCRUBBER WASTE WATER	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000
NOX SUPPRESSION	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000
TOTAL LAND ACRES/100MWE	86.51	.00	.00	31.45	89.55	80.42	91.79	89.07
MAIN PLANT	19.16	.00	.00	19.15	19.10	19.17	.00	19.13
DISPOSAL LAND	33.12	.00	.00	31.69	29.38	41.33	.00	37.39
LAND FOR ACCESS RR	32.53	.00	.00	32.58	30.00	40.72	.00	32.58
PARAMETRIC POINT	17	18	19	20	21	22	23	24
COAL, LB/KW-HR	.00000	.31178	.00000	.00000	.00000	.00000	.00000	.00000
SORBANT OR SEED, LB/KW-HR	.00000	.48242	.00000	.00000	.00000	.00000	.00000	.00000
TOTAL WATER, GAL/KW-HR	.000	.343	.000	.000	.000	.000	.000	.000
COOLING WATER	.000	.343	.000	.000	.000	.000	.000	.000
GASIFIER PROCESS H2O	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000
CONDENSATE MAKE UP	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000
WASTE HANDLING SLURRY	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000
SCRUBBER WASTE WATER	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000
NOX SUPPRESSION	.00000	.00000	.00000	.00000	.00000	.00000	.00000	.00000
TOTAL LAND ACRES/100MWE	.00	130.45	.00	.00	.00	.00	.00	.00
MAIN PLANT	.00	19.73	.00	.00	.00	.00	.00	.00
DISPOSAL LAND	.00	79.82	.00	.00	.00	.00	.00	.00
LAND FOR ACCESS RR	.00	31.90	.00	.00	.00	.00	.00	.00

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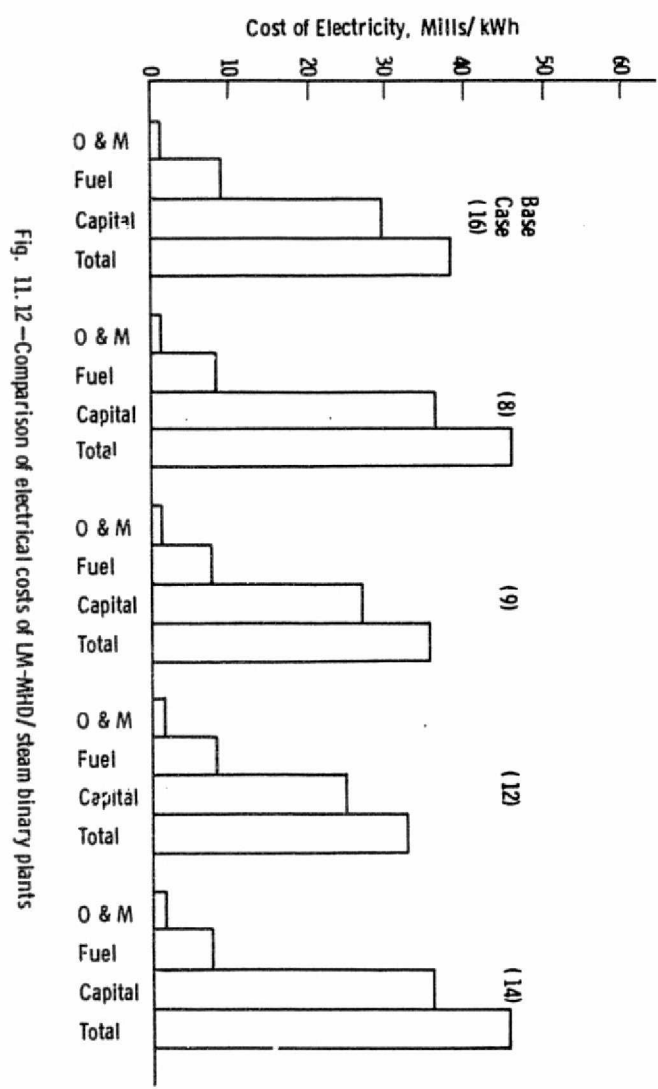


Fig. 11.12 - Comparison of electrical costs of LM-MHD/ steam binary plants

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operating and maintenance costs are relatively small for all cases, compared to either the capital equipment costs or fuel costs. The fuel costs range from about 1/4 to 1/3 of the capital costs. For the comparisons shown, fuel costs were taken at \$0.805/GJ ( $\$0.85/10^6$  Btu).

The effect of the fuel costs on the total costs of electricity is shown in Figure 11.13 for the five plants. A comparison of these fuel cost curves indicates a difference in the slope of the cost line for various parametric cases. This difference in the slopes is due to differences in power plant efficiency. The higher the power plant efficiency, the less dependent is the cost of electricity on fuel cost.

The effect of power plant efficiency on relative fuel costs is shown in Figure 11.14. In this figure the fuel costs relative to the base case are plotted with respect to power plant efficiency for the six LM-MHD steam binary plants. Each of these power plants represents a different aspect of LM-MHD technology that requires development. Developments in duct and component efficiencies as well as higher duct inlet temperatures are indicated, and both effects are seen to act to reduce fuel costs. The combined effect of the most optimistic component performance with the high duct inlet temperature would be about a 14% reduction in the base case fuel costs, i.e., from 2.39 to 2.87 mills/MJ (8.61 to 7.46 mills/kWh).

Fuel conservation is an important consideration in today's economy, but fuel costs are only one aspect that must be considered. The total cost of electricity is still the deciding criterion. As such, a comparison of total costs of electricity for the various LM-MHD power plants as a function of power plant efficiency is shown in Figure 11.15. The numbers in the circles in the figure correspond to the respective parametric points. Also indicated beside the points are the technology achievements assumed in developing each of the study points. Thus, in assessing this figure, it can be seen, for example, that increasing the duct inlet temperature over the base case (Point 16 compared to Point 8) results in increased plant efficiency but at significantly increased plant costs. Some of this cost increase is recoverable if higher duct efficiencies are assumed (i.e.,



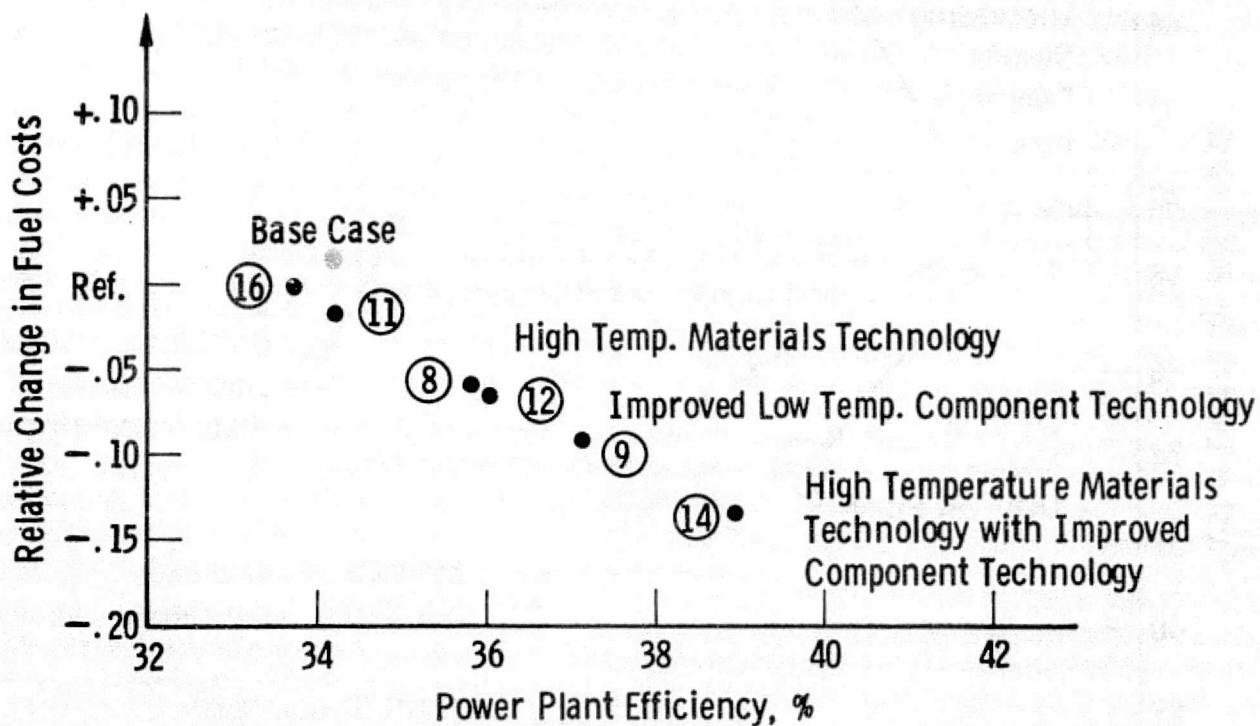


Fig. 11.14 — Comparison of fuel cost and power plant efficiency for different LMMHD/ steam binary plants

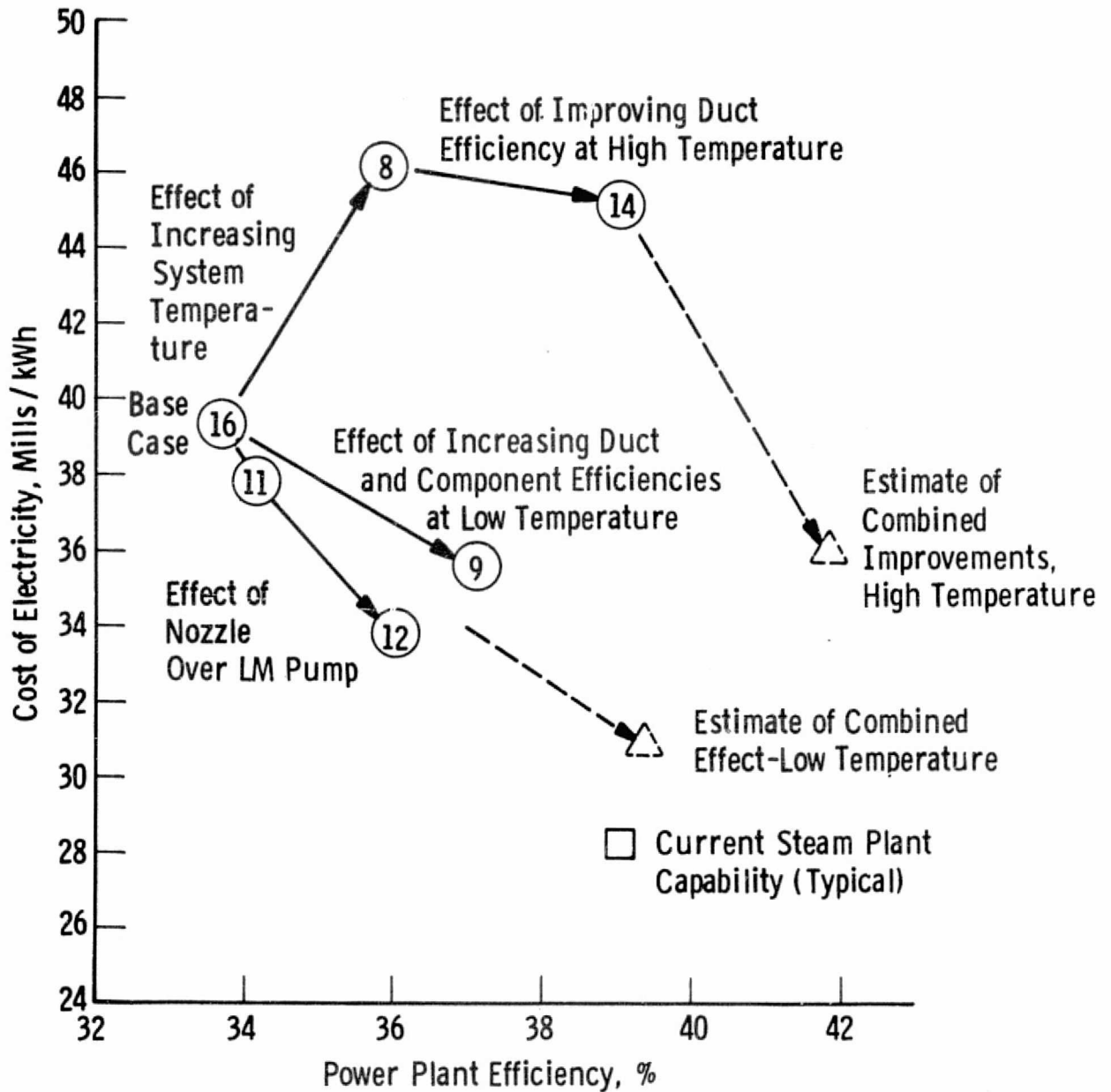


Fig. 11.15—The effect of technology advancement on LM-MHD/ steam binary power plant performance and cost

in going from Point 8 to Point 14). In a similar manner, if we assume only improved duct and component efficiency over the base case (no increase in temperature) the results are increased plant efficiencies and decreased costs. The most predominant effect here is to replace the costly liquid-metal mechanical pumps with a highly efficient two-phase nozzle diffuser.

The dotted lines in the figure show how the trends in this analysis might be extrapolated to account for the combined effects of both higher component and duct efficiencies and the utilization of a liquid-metal two-phase nozzle diffuser. These combined effects are shown for both the low and high duct inlet temperatures. Comparing these two cases then, would suggest that the high-temperature duct does not lead to the most cost effective power plant. The cost and performance for the high-temperature duct with combined improvements are 10.0 mills/MJ (36 mills/kWh) at about 42%, respectively. The low-temperature case is 8.61 mills/MJ (31 mills/kWh) at 39%. A typical value for current steam power plants is shown for comparison.

The major conclusion that could be inferred from this figure is that any near term LM-MHD development effort might best be spent improving and demonstrating the duct and component efficiencies for the low-temperature application. Development of a highly efficient, two-phase nozzle diffuser would impact significantly on the economic viability of the plant.

It should be noted that the comparison made in Figure 11.15 is based on a fuel cost of \$0.805/GJ ( $\$0.85/10^6$  Btu). Should significantly higher fuel cost be incurred, the difference between the costs of the low- and high-temperature case diminishes. For example, at the upper range of coal costs considered in this study [ $\$2.37/\text{GJ}$  ( $\$2.50/10^6$  Btu)], the cost of electricity for the two cases would become comparable, but the high-temperature case would hold an efficiency advantage. Care must be exercised, therefore, in evaluating plant costs when both temperature-dependent material costs as well as fuel costs are involved.

### 11.6.3 All MHD Power Plants

Two parametric cases were studied that represent all MHD power plants. Point 4 involved an open-cycle MHD topping the LM-MHD. The

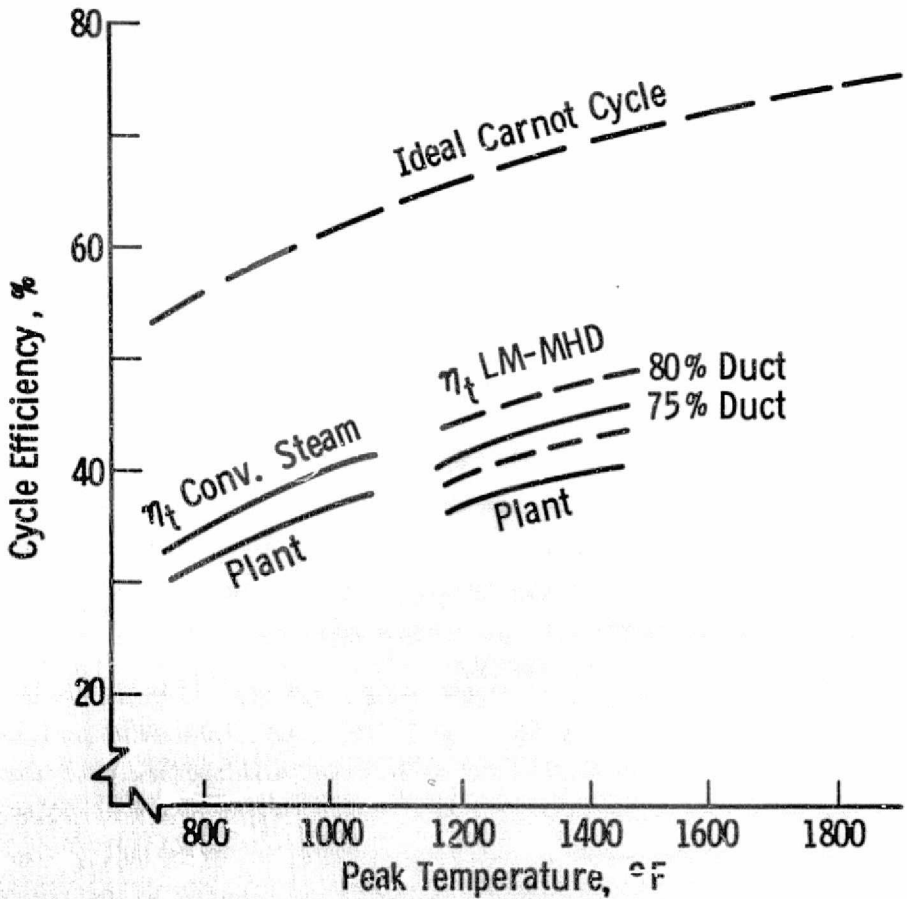


Fig. 11. 16—Efficiency overview LM-MHD system

cost of electricity was about 15.28 mills/MJ (55 mills/kWh), with the major cost factor again being the capital equipment associated with the LM-MHD system.

The second all-MHD plant, Point 13, was an all LM-MHD system producing 1005 MWe net but requiring over 3800 MWe gross. The power costs associated with converting 3800 MWe from dc to ac and the high system flow rates resulted in prohibitive power generation costs. The cost of electricity was nearly 22.2 mills/MJ (80 mills/kWh).

#### 11.7 Conclusions and Recommendations

Results of the study show that of the near term cases studied [near term being dictated by the lower temperature sodium technology - 922°K (1200°F)], none would appear to offer attractive plant performance. The potential of the direct coal-fired LM-MHD steam binary system exists with the further development of liquid-metal component technology and demonstration of high MHD duct efficiencies. Higher duct inlet temperatures, 1088°K (1500°F) lead to higher plant efficiency but not necessarily to lower electrical costs. At 1088°K (1500°F), power plant efficiencies of about 43% are attainable contingent on achieving MHD generator efficiencies of 80% and assuming marginal improvements in steam plant technology. Further studies are recommended to define and evaluate the duct variables and trade-off that lead to high duct efficiencies.

An overview of the potential performance of the LM-MHD binary systems is depicted in Figure 11.16. The operating regime of the LM-MHD system is compared to conventional steam plants and the ideal Carnot cycle as a function of peak conversion temperature. The purpose of the LM-MHD topping cycle is to extend the operating temperature range and efficiency of the power plant in a manner predicted by the ideal Carnot cycle. Real effects, however, can limit the attainment of this increased efficiency. As noted, no penalty on plant performance appears to exist for the LM-MHD binary plant with an 80% MHD duct efficiency. At 75% duct efficiency, however, an approximate 373°K (671°F) temperature penalty is incurred

in order to obtain increased system performance over the conventional steam plant. In other words, a 978°K (1300°F) LM-MHD system appears to be the threshold for improved plant performance for this case.

Since the conventional steam plant operation is currently in the range of 811°K (1000°F), the conclusion inferred from Figure 11.16 is that there is a potential gain in plant performance by going to a LM-MHD steam binary system with peak temperatures above 922°K (1200°F) if the LM-MHD system components can be made to meet the assumed performance levels. The major cost factors that limit the economic potential of the LM-MHD/steam binary plant are the large costs for power conversion, heat transfer equipment, and liquid-metal pumping. The lowest range of electrical costs determined for the cases studied was about 9.44 mills/MJ (34 mills/kWh), which could be reduced to about 8.33 mills/MJ (30 mills/kWh) with improvements in LM-MHD components. In all cases studied, however, the costing was largely uncertain, since experience in design and fabrication of LM-MHD components is not available.

It is expected that through innovative design and with the demonstration of improved high-temperature metals technology, heat transfer equipment costs could be reduced. Reduced power conversion costs are associated with still higher duct efficiencies. Development of a high performance direct ac generator (i.e., slug-flow type) would eliminate the need for much of the expensive inverter equipment. The demonstration of a large-scale, high-performance, two-phase flow nozzle diffuser would both reduce costs and improve cycle performance. A cost advantage is achieved in both the liquid-metal pump hardware and in the power conversion equipment.

The results of this study are believed to be adequate for the needed system assessments on a relative basis, but specific conclusions on performance and cost should be revised with increased understanding of large-scale LM-MHD component design, performance, and costs. Because of the complexity in interfacing LM-MHD cycles with Rankine-type cycles, further studies are required to optimize plant performance with respect to various cost factors. Of particular importance are the trade-offs

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between the MHD duct operating temperatures with respect to material cost considerations and fuel cost.

The studies made on the two all-MHD systems must be considered preliminary and were not adequate to determine the potential for this power plant. Both plants showed low efficiency and high cost, in part due to the modest assumptions made with respect to the LM-MHD component and duct efficiencies. These studies, however, should provide a guide for identifying major cost considerations and determining cycle configurations that could lead to improved plant performance.

It is recommended that no full-scale commercial plant design study of the LM-MHD system be undertaken in Task II of the ECAS program. Other advanced conversion systems under study in Task I would appear to offer the potential for higher performance and lower costs with less uncertainty associated with the results. Efforts on LM-MHD should, however, be encouraged in order to develop sufficient data to predict reliably full-scale MHD generator efficiencies as well as associated component performance.

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## Appendix A 11.1

### LIQUID-METAL MHD TWO-PHASE FLOW AND PRESSURE LOSS CONSIDERATIONS

The two-phase flow components of the LM-MHD System are the mixer, the MHD duct, the nozzle, and the separator.

#### A 11.1.1 Mixer

The mixer provides a homogeneous (bubbly) flow into the MHD duct. Within the mixer, the two fluids (argon and sodium) are brought together and mixed to form the desired flow. The problems associated with the mixer are the evaluation of the pressure loss incurred and the design of a system for minimum pressure loss. Although a significant technology base exists for fluid mixer designs, the two-phase (gas-liquid metal) systems of the component size required for this study makes it a development item. Indeed, the need for homogeneous (bubbly) flow with low pressure loss has led to the need for a preliminary conceptual design.

Preliminary analysis shows that, based on the desired MHD duct inlet flow void fraction (65%), the two-phase flow would naturally tend toward a dispersed-type flow pattern (liquid droplets entrained in the gas). This observation is based on established correlations for adiabatic, horizontal, two-phase, two-component flow for air-water systems. Thus, initiating and maintaining a bubbly flow will be difficult. Factors to be considered in establishing bubbly flow include the mechanism of bubble generation, bubble size, and bubble influence on the dynamics of the mixture.

The usual method for introducing a gas into a liquid medium is through an orifice or porous plate. Depending on the gas velocity, the gas can be introduced as discrete bubbles that are formed at the orifice and subsequently break off or as a gas jet which eventually breaks into individual bubbles. When discrete bubbles are formed at the orifice,

the equilibrium bubble size (break-off size) is determined by balancing buoyance, surface tension, inertia, and viscous forces.

The pressure drop across an orifice where bubbles are being formed is given approximately by

$$\Delta p = \frac{2\sigma}{R}$$

where R is the bubble radius at break-off and  $\sigma$  is the surface tension. For sodium/argon, where  $\sigma \approx 0.191$  N/m (0.0131 lb/ft), the following table is constructed.

Table A 11.1.1

Bubble Radius vs. Orifice Pressure Drop

<u>Bubble Radius, in</u>	<u><math>\Delta p</math> psi</u>	
0.0001	22	} Range for porous plates
0.001	2.2	
0.01	0.22	
0.10	0.022	} Range for orifices
0.25	0.0087	
0.50	0.0044	

Assuming that bubble size is on the same order of magnitude as the orifice radius, the above table indicates the range of pressure drops that could be expected under ideal conditions where dynamic and interference effects are neglected.

The criteria of a jet issuing from an orifice (as opposed to discrete bubble formation) are given by the dimensional Equation A 11.1.1 (Reference 11.9).

$$\frac{V_g \sqrt{\rho_g}}{[g_c g \sigma (\rho_f - \rho_g)]^{1/4}} = 1.25 \left[ \frac{\sigma}{(\rho_f - \rho_g) R_o^2} \right] \quad (\text{A 11.1.1})$$

where  $V_g$  is the gas jet velocity,  $\rho_f$  and  $g_c$  are the liquid and gaseous component densities,  $R_o$  is the orifice diameter, and  $g$  is the gravitational constant, or, for the sodium-argon system, for example,

$$0.317 V_g \geq \frac{0.035 \times 10^{-2}}{R_o}$$

Where  $R_o$  is in ft and  $V_g$  is in ft/s.

Assuming  $V_g = 100$  ft/s.

$$0.325 V_g \geq \frac{0.0208}{R_o}$$

$$R_o \geq 0.0077 \text{ in}$$

Thus, for an orifice radius of 0.195 mm (0.0077 in) or larger the argon gas will enter as a jet stream; below this orifice, discrete bubbles will form.

The pressure loss in the mixer can be determined from a momentum balance across the system. The gas-side pressure requirement is obtained by considering the details of the method of injection. If the bubbles are injected so that discrete bubbles form at the wall of the mixer, the gas-side pressure drop is given as

$$\Delta p = \frac{2\sigma}{R_o}$$

where  $R_o$  is the radius of the orifice. Pressure losses for various orifice sizes are given in Table A 11.1.1. If, on the other hand, the

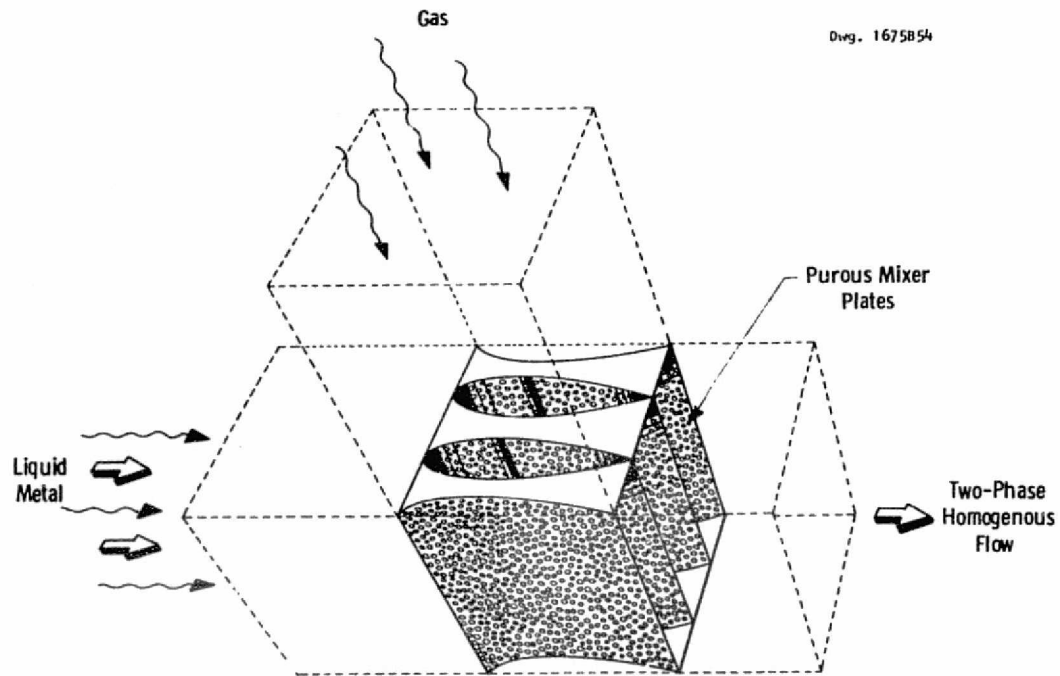


Fig. A 11.1.1—Low pressure loss-L. M. MHD liquid-gas mixer, conceptual design

design is such that the gas enter the liquid as a jet stream (subsequently breaking into bubbles), the pressure loss is given by the usual orifice equation:

$$(P_g - P_{mix}) = \frac{1}{2} \frac{\rho_g V_g^2}{g_c} + K \frac{\rho_g V_g^2}{2 g_c} \quad (\text{A 11.1.2})$$

For worst case conditions, K can be taken as 1. Taking  $V_g$  at 30.48 m/s (100 ft/s), the pressure loss across the orifice is 40.06 kPa (5.81 psi) ( $\Delta p/p = 0.0048$ ). The preferred method of mixing the gas and liquid would be by formation of a gas jet stream that sufficiently penetrates the liquid stream. With the cases where discrete bubbles form at the wall, there would be a high likelihood of bubbles agglomerating within the liquid-wall boundary layer which could lead to an undesirable slug flow development.

The design of the mixer can be based on two cases:

- Accelerating the liquid metal prior to its entry into the mixer to the required duct velocity and designing the mixer to maintain a constant mixture velocity
- Designing the mixer for constant area so that the gas-liquid mixture is accelerated to the desired duct velocity at the exit of the mixer.

An advantage may exist with the first approach, since this design may enhance the stability of the bubbly flow. Figure A 11.1.1 shows a conceptual design of one-mixer system where gas is injected into the liquid through air-foil shaped plates. This design leads to low friction losses in the mixer (approximately 2 psi assumes an effective  $\frac{L}{D}$  of 20); and, accordingly, this friction loss was neglected compared to the losses associated with other major components.

#### A 11.1.2 MHD Duct

Two-phase, two-component LM-MHD generators have been tested at low temperatures in small-scale models by several groups using NaK and nitrogen. Duct efficiencies of about 50% have been reported (References 11.1, 11.2, and 11.3) with recent developments in identifying and controlling boundary layer, shunt losses, and end-loss attenuation two-phase theory and experimental data have been brought into closer agreement. When applied to large-scale systems, these models predict MHD duct conversion efficiencies of around 75%.

Additional experiments in the USSR and West Germany (References 11.10 and 11.11) have shown good performance without degradation at higher temperature [ $\sim 800^\circ\text{K}$  ( $980^\circ\text{F}$ )]. The principal performance parameters appear to indicate as good or better performance on large-scale systems and would lead one to propose maximizing duct size and plant power levels. These performance parameters do not, however, reflect the real-life limitations imposed by structural and physical system requirements. These include structural designs capable of operation at temperature and high pressures [10.13 MPa (100 atm)] and the transmission by conductors of large amperage currents at low voltages from the ducts to inverter systems. Such limitations as these result in size and arrangement constraints on the MHD ducts.

The nature of the two-phase flow within the MHD duct has been described as churn turbulent-bubble flow. Because of the complex nature of this flow, in part due to the uncertain effects of the interaction between slip and electrodynamic forces, an analysis of this flow field cannot be made in closed form. Empirical expressions have been developed, however, that attempt to account for the major flow phenomena. As indicated above, collaboration of these models on small systems have been successful to the extent that experimental MHD duct performances can be predicted over limited operation conditions. The stability of the bubble regime within the MHD duct in a large system remains uncertain. The electrodynamic force retarding the motion of the liquid, coupled

with a gas transpiration (gas injection to circumvent electrical shunting) along the duct walls may act to retard formation of the stable, dispersed annular flow regime. Details of the calculational model used to predict MHD duct performance in this study are described in detail in Appendix A 11-2.

#### A 11.1.3 Nozzle

After passing through the MHD duct, the flows are separated into two fluid streams (sodium/argon), and the liquid pumped back to the inlet pressure conditions. The technical complexity and practicability of these requirements are severe. The large volume of liquid metal that must be pumped [ $63.09 \text{ m}^3/\text{s}$  ( $10^6 \text{ gpm}$ )] far exceed current state of the art in liquid-metal pump sizes. Projected mechanical pump sizes for commercial fast-breeder reactor plants are on the order of  $3.785 \text{ m}^3/\text{s}$  (60,000 gpm).

An alternative to using a liquid-metal mechanical pump would be to impart sufficient dynamic head to the flowing liquid metal (from the gases) through the two-phase nozzle that is connected to the exit of the MHD duct. The objective then would be to recover this dynamic head (after the two fluids are separated) in a liquid diffuser. The hydrodynamics of such an approach are complex. The requirements on the acceleration of the liquid phase by the gas in the nozzle section would be highly dependent on flow regime. The flow tends toward annular or dispersed annular regime. In annular flow, the liquid metal flows along the containing walls, and viscous effects and phase slip would severely limit liquid flow velocity. With dispersed flow, the liquid exists as droplets entrained in the gas phase. Although the phases are close coupled so that the required dynamic head could be imparted in the nozzle, the losses that would be incurred in separating the phases would be greatly increased. The problem is further complicated, since the required velocities in the nozzle are such that sonic flow conditions in the two-phase system could result. The required liquid velocity corresponding to an 8.274 MPa (1200 psi) dynamic head is 143.2 m/s (470 ft/s). Sonic flow conditions correspond to about 152 m/s (500 ft/s).

The affect on the flow of achieving sonic and near sonic flow conditions is to increase system pressure losses, increase erosion of materials (because of high velocity), and enhance structural problems resulting from vibrations and "water hammer" effects.

A second alternative to conventional liquid-metal pumps is an electromagnetic (EM) pump. The concept would employ a return duct within the same magnetic field developed for the MHD duct. The power required to drive the EM pump could be shunted from one or more of the adjacent MHD power ducts. The practicability of building such a pump and its resulting performance are uncertain. State of the art on such systems is for comparatively small scale, 0.00012 to 0.315 m<sup>3</sup>/s (2 to 5,000 gpm). Developmental costs for large EM pump systems, however, would not be expected to be as exorbitant as the LM pumps, since the required magnet and ducting are scalable. A preliminary design and cost of one 15.77 m<sup>3</sup>/s (250,000 gpm) EM pump is described in Appendix A 11.5.

#### A 11.1.4 Separator

The gas/liquid separator system must effectively separate the two fluids with minimum pressure loss. Should the separation be ineffective, carry-over of liquid metal with the gas will occur. This may be in the form of an oxidized aerosol that could carry for some distance in the piping system. Such aerosols may present problems with rotary seals and blade erosion in the gas compressor system. Experimental data on small-scale impingement-type, liquid metal-gas separator systems indicate 95 to 99% effectiveness.

Large pressure losses in the separator degrade system performance. This is particularly true where high velocity heads are imparted to the liquid metal by the gas and subsequently recovered in a liquid-metal diffuser. Again, with small-scale systems, 10% pressure losses are common.



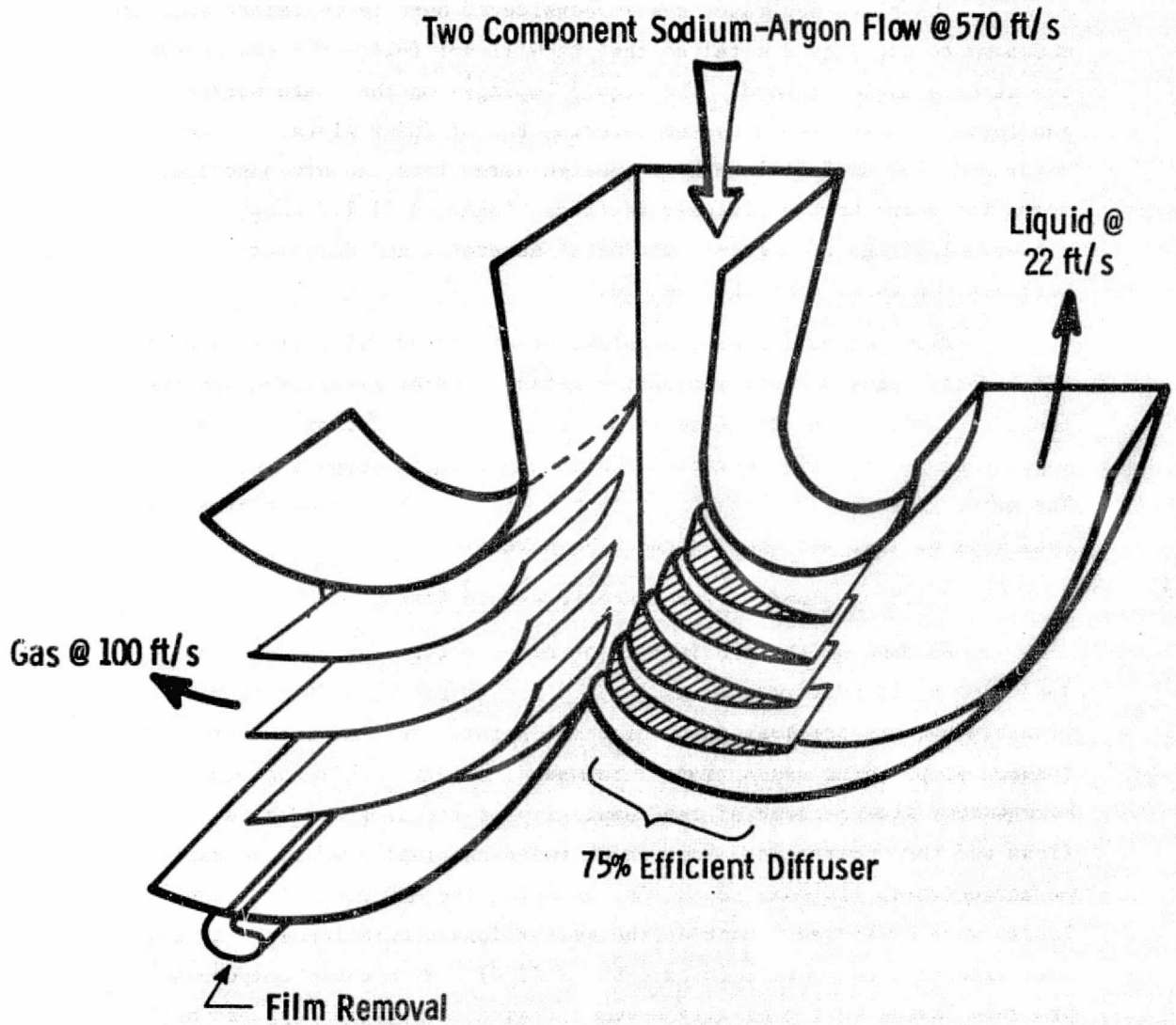
The plate-separator system considered here is to impart sufficient momentum to the liquid metal so that it will not follow the gas through the turning angle or bend. The liquid impinges on the plate surface and forms a fully liquid system entering the diffuser slots. Plate angle and flow surface length are design parameters, as are also slot sizes for entry to the diffuser section. Figure A 11.1.2 shows a conceptual design of a gas-liquid metal separator and diffuser that utilizes the above described method.

Flow regime (bubble, annular, or dispersed) will greatly affect the effectiveness of this separation method. Other separation schemes include decelerating the flow to allow gravity separation or direct impingement of the liquid-metal droplets in a radial-type turbine. The major disadvantage with these systems is that the dynamic pressure head will be lost and erosion could be severe.

#### A 11.1.5 Evaluation for System Pressure Losses

Because of the sensitivity of Brayton-type cycles to pressure losses on cycle performance, care was taken to assure a fair (and probably optimistic) evaluation of this factor. To evaluate these losses, simplifying assumptions were made (namely, one-dimensional homogeneous flow) because of the complexity of treating two-phase flows and the uncertainty about which two-phase region would actually be encountered. In some cases—for example, the MHD duct—friction losses were neglected. Most of the system losses were incurred in the heat exchanger components (single-phase flow). With these components pressure losses were generally assumed (based on experience), and heat exchanger designs were developed that would satisfy the constraints.

Table A 11.1.2 summarizes the basis for evaluation and pressure loss values that were used in evaluating the LM-MHD cycle.



Separator/Diffuser Concept (Side Removed)

Fig. A 11.1.2—Conceptual design of nozzle

Table A 11.1.2

## Summary of the LM-MHD Component Pressure Losses

Component	Basis for Evaluation	Loss, psi	Pressure Loss as Fraction of Duct Inlet Pressure
Mixer			
Gas	Orifice pressure loss	5.5	0.0046
Mixture	Homogenous 2 $\phi$ flow friction loss	4.5	0.0075
MHD Duct	Friction losses neglected	-	
Nozzle - Mixture	95% efficiency	19.4	0.016
Separator			
Gas & Diffuser	Friction loss	6	0.005
Liquid	Loss of the velocity component corresponding to deflection angle	15	0.012
Steam Generator	Designed	20	0.017
Primary Heat Exchanger			
Gas	Single-phase friction loss	50	0.042
Liquid	Single-phase friction loss	65	0.054
Recuperators/ Intercoolers and Air Preheaters	2% of component inlet pressure	As required	0.02

## Appendix A 11.2

### DESCRIPTION OF THE LM MHD DUCT AND CYCLE COMPUTER MODEL

A computer program was written for rapid evaluation of LM-MHD cycles. This report describes the basic assumptions and analytical methods incorporated.

In the cycle discussed here a gaseous phase is dispersed in a liquid metal at high pressure and temperature. This mixture expands through an MHD duct in which the liquid metal provides the electrical conduction path. The two components are separated downstream of the duct, and the reject heat is removed from the gas stream. The two phases are separately pumped back to duct inlet pressure and heat added to one or both streams prior to remixing.

#### A 11.2.1 Continuity Relations

Three different measures of gas fraction in the mixture are used which we define as follows:

$w_g$  = fraction of total mass flow rate as gas

$w_l$  = fraction of total mass flow rate as liquid

$\phi_g$  = fraction of total volume flow as gas

$\phi_l$  = fraction of total volume flow as liquid

$\beta_g$  = fraction of cross-sectional area occupied by gas

$\beta_l$  = fraction of cross-sectional occupied by liquid.

In addition, we let

$U_g$  = gas velocity

$U_l$  = liquid velocity

$v_g$  = gas specific volume

$v_l$  = liquid specific volume.

Continuity demands the following identities:

$$w_g = \frac{\frac{\phi_g}{v_g}}{\frac{\phi_g}{v_g} + \frac{\phi_l}{v_l}} \quad (\text{A 11.2.1})$$

$$\phi_g = \frac{w_g v_g}{w_g v_g + w_l v_l} \quad (\text{A 11.2.2})$$

$$\phi_g = \frac{\beta_g U_g}{\beta_g U_g + \beta_l U_l} \quad (\text{A 11.2.3})$$

$$\beta_g = \frac{\frac{\phi_g}{U_g}}{\frac{\phi_g}{U_g} + \frac{\phi_l}{U_l}} \quad (\text{A 11.2.4})$$

### A 11.2.2 Thermodynamic Relations

The ideal adiabatic expansion (or compression) of a gas and liquid mixture in thermal equilibrium obeys the same laws as does a pure gas phase, with slight modification. Assuming, in addition, that the

gas is a perfect gas, and that the liquid has constant properties and negligible vapor pressure, then the following relations hold:

$$Pv_g = RT \quad (\text{A } 11.2.5)$$

$$w_g PdV_g + w_g C_{vg} dT + w_l C_l dT = 0 \quad (\text{A } 11.2.6)$$

$$C_{pg} = C_{vg} + R \quad (\text{A } 11.2.7)$$

If we define

$$\gamma = \frac{C_{vg} w_g + R w_g + C_l w_l}{C_{vg} w_g + C_l w_l} \quad (\text{A } 11.2.8)$$

as the ratio of weighted mean specific heats of the mixture, then the integral expression becomes:

$$\frac{T_2}{T_1} = \left( \frac{P_2}{P_1} \right)^{\frac{\gamma-1}{\gamma}} \quad (\text{A } 11.2.9)$$

$$\frac{v_{g2}}{v_{g1}} = \left( \frac{P_2}{P_1} \right)^{-\frac{1}{\gamma}} \quad (\text{A } 11.2.10)$$

These expressions must be modified to account for entropy changes that will occur in a real situation. Thus, as in gas turbine practice, we may define isentropic or polytropic expansion efficiencies such that:

$$T_1 - T_2 = T_1 \eta_t \left[ 1 - \left( \frac{P_2}{P_1} \right)^{\frac{\gamma-1}{\gamma}} \right] \quad (\text{A } 11.2.11)$$

$$T_1 - T_2 = T_1 \left[ 1 - \left( \frac{P_2}{P_1} \right)^{\frac{(\gamma-1)\eta_p}{\gamma}} \right] \quad (\text{A } 11.2.12)$$

The small stage efficiency in the MHD duct, however, will normally vary significantly during the expansion. This fact requires a numerical integration of the differential form of the above equations. This may take the form:

$$dT = \frac{(\gamma-1)\eta_p}{\gamma} \frac{T}{P} dP, \quad (\text{A } 11.2.13)$$

$$dP = \frac{P}{T} \frac{\gamma}{(\gamma-1)\eta_p} dT, \text{ or} \quad (\text{A } 11.2.14)$$

$$dP = \frac{\gamma}{\gamma(\eta_p-1)-\eta_p} \frac{P}{v_g} dv_g, \quad (\text{A } 11.2.15)$$

depending on whether one wished to integrate between pressure, temperature, or specific volume (void fraction) limits.

### A 11.2.3 Efficiency Estimates

The performance of the IM-MHD two-component, two-phase cycle, as with any Brayton-like cycle is acutely sensitive to irreversibilities in the individual components. The expansion efficiency is of particular concern. The following sections are thought to include the most significant expansion losses.

A 11.2.3.1 Drag Losses Due to Slip between the Liquid and Gas

The liquid and gas are equally subject to forces arising from a pressure gradient in the direction of flow. These forces are countered in the liquid metal by body forces of electromagnetic origin implicit in the conversion to electrical energy. Since the pressure forces on the gas are resisted only by drag between the liquid and gas, it necessarily follows that the gas velocity will be greater than the liquid velocity and that pressure and kinetic energy will be degraded back into thermal energy by effects of drag forces and relative velocity.

The momentum equations for the gas and liquid components are

$$-\frac{dP}{dx} + D_g - \frac{d}{dx} (\rho_g U_g^2) = 0 \quad (\text{A 11.2.16})$$

$$-\frac{dP}{dx} + D_l + f_l - \frac{d}{dx} (\rho_l U_l^2) = 0, \quad (\text{A 11.2.17})$$

where  $D_g$  and  $D_l$  are respectively the drag in units of force per unit volume in gas and liquid, all quantities taken as positive in the direction of flow.

Note:

$$D_g \beta_g = -D_l (1 - \beta_g). \quad (\text{A 11.2.18})$$

For brevity let

$$R_g = \frac{d}{dx} (\rho_g U_g^2) \quad (\text{A 11.2.19})$$

$$R_l = \frac{d}{dx} (\rho_l U_l^2). \quad (\text{A 11.2.20})$$



Then

$$D_g = \frac{dP}{dx} + R_g \quad (\text{A } 11.2.21)$$

$$f_l = \frac{dP}{dx} - D_l + R_l = \frac{dP}{dx} + \frac{\beta_g}{1-\beta_g} \left( \frac{dP}{dx} + R_g \right) + R_l$$

$$f_l = \frac{\frac{dP}{dx} + \beta_g R_g + (1-\beta_g) R_l}{1-\beta_g} \quad (\text{A } 11.2.22)$$

The mechanical energy converted to electrical (other losses omitted) per unit volume of mixture per unit time is

$$\epsilon_c = -f_l U_l (1-\beta_g) \quad (\text{A } 11.2.23)$$

This mechanical energy reconverted to thermal is:

$$\epsilon_d = -D_g \beta_g (U_g - U_l) \quad (\text{A } 11.2.24)$$

The efficiency reduction factor for slip is then

$$\eta_s = \frac{\epsilon_c}{\epsilon_c + \epsilon_d} = \frac{-f_l U_l (1-\beta_g)}{-f_l U_l (1-\beta_g) - D_g \beta_g (U_g - U_l)} \quad (\text{A } 11.2.25)$$

$$\eta_s = \frac{-U_l [R_l (1-\beta_g) + R_g \beta_g + \frac{dP}{dx}]}{-U_l R_l (1-\beta_g) - U_g R_g \beta_g - U_l (1-\beta_g) \frac{dP}{dx} - U_g \beta_g \frac{dP}{dx}}$$

$$\eta_s = \frac{[R_l (1-\beta_g) + R_g \beta_g + \frac{dP}{dx}]}{R_l (1-\beta_g) + (1-\beta_g) \frac{dP}{dx} + S(R_g \beta_g + \beta_g \frac{dP}{dx})} \quad (\text{A } 11.2.26)$$

where

$$S = \frac{U_g}{U_l}$$

It is noted that quantities  $S$ ,  $\beta$ , and  $dP/dx$  dominate this expression. In order to evaluate  $\eta_g$  the slip ratio,  $S$ , must be determined, depending in turn on the relationship between drag force and relative velocity. Argonne National Laboratory personnel have suggested the formula

$$D_g = -C_D \rho_l (U_g - U_l)^2 (1 - \beta_g)^3 \quad (\text{A } 11.2.28)$$

based on experimental data by Zuber et al. (References 11.12 and 11.13).

It is now assumed that for a Faraday generator, the magnetic flux, the duct width normal to the magnetic field, and the liquid velocity will be held constant, as far as is possible. This will tend to hold a uniform, open-circuit voltage.

From Equation A 11.2.16

$$-\frac{dP}{dx} - C_D \rho_l (U_g - U_l)^2 (1 - \beta_g)^3 - U_g^2 \frac{d\rho_g}{dx} - 2 \rho_g U_g \frac{dU_g}{dx} = 0$$

$$2\rho_g S \frac{dS}{dx} + C_D \rho_l (S - 1)^2 (1 - \beta_g)^3 + S^2 \frac{d\rho_g}{dx} + \frac{1}{U_l^2} \frac{dP}{dx} = 0 \quad (\text{A } 11.2.29)$$

An examination of the numerical values of the terms of this equation show that the drag force and pressure gradient terms are highly dominating. This justifies omitting the term involving the gradient in slip and calculating slip, on a quasi steady-state basis, from the local gradients in pressure and density.

This slip value may be substituted into Equation A 11.2.26 to estimate the slip contribution to local duct efficiency losses. As applied here

$$R_2 = 0$$

consistent with axially uniform liquid velocity.

#### A 11.2.3.2 Ohmic Losses from Nonuniform Induced EMF and Fluid Shear Losses

These losses are closely related, and it does not seem possible in the general case to decouple the two contributors except for end losses.

In ordinary pipe flow the total pressure gradient along the length of the pipe acts primarily to overcome fluid shear stresses. In an MHD generator it is intended that the pressure gradient react primarily against electromagnetic body forces, thereby resulting in useful energy conversion, with fluid shear stresses kept reasonably small. It is, thus, not expected that the velocity distribution in an MHD duct will closely resemble that usually encountered in pipe flow. Any nonuniformity in the magnetic field -- in the magnetic field direction -- will, for example, result in a corresponding nonuniformity in the electric field that will tend to increase the ohmic losses. It will also result in velocity gradients that will increase fluid friction losses and feedback into the electric field distribution. Velocity gradients at the boundary resulting from wall drag will similarly interact with the voltage and current fields.

This fairly complex source of efficiency loss has not been treated in detail. It has simply been assumed that the voltage and velocity fields are uniform. The result is an efficiency factor of  $E/E_0$  that is the working voltage over the open-circuit voltage, as applied to the MHD duct proper.

At each end of the duct the induced voltage drops with the magnetic field and no longer counters the applied voltage. This results in reverse current flow and loss of electrical energy in ohmic heating. The effective resistance of an infinitely long approach with a sharp magnetic field drop was calculated by Sutton (Reference 11.13) as

$$R_e = \frac{\pi \delta}{2W\mu n(2)} \quad (\text{A 11.2.30})$$

The ohmic losses can then be calculated as  $E^2/R_e$  for each end of the duct. It has been shown experimentally at Argon that these losses can be reduced by approximately an order of magnitude by appropriately tapering the magnetic field.

#### A 11.2.3.3 Departure from Thermal Equilibrium

As noted above, the ideal cycle is predicted on thermal equilibrium between the liquid and gas. In actuality, the expansion work performed by the gas component results, for the most part, from the thermal energy of the liquid. Thus, there will necessarily be a temperature drop across the liquid-gas interface, whose value depends on the required rate of heat transfer and the thermal resistance. It is expected that the gas will be dispersed into a large number of small bubbles with sufficient surface area that the temperature difference will be small. For present purposes this source of irreversibility is not studied.

#### A 11.2.3.4 Thermal Losses through the Duct Wall

These constitute a direct source of energy loss and may be estimated from the overall thermal resistance from the inside of the duct wall. For large MHD ducts these are expected to be small and have not been studied.

#### a 11.2.4 Electromagnetic Relationships

It is assumed that the current through the duct does not affect the magnetic field. This can be approximated by proper arrangement of the conduction path outside the duct so that no net magnetic flux is contributed by the duct current. Then the open-circuit voltage is

$$E_o = B U_l t \quad (\text{A } 11.2.31)$$

where  $E_o$  is the volts

$B$  is magnetic flux, T

$U_l$  is the liquid velocity, m/s

$t$  is duct thickness, m.

The current density,  $j$ , is related to the working voltage,  $E$ , by Equation A 11.2.32:

$$j = \frac{E_o - E}{\delta t} \quad (\text{A } 11.2.32)$$

where  $\delta$  is the mixture resistivity.

Mixture resistivity is estimated as

$$\delta = \frac{E_o (3.8 \beta)}{\kappa} \quad (\text{A } 11.2.33)$$

where  $\kappa$  is the pure liquid conductivity (Reference 11.12).

The linear electrical power density is given by

$$\frac{dp}{dx} = E j W \quad (\text{A } 11.2.34)$$

By equating the electrical power density to the gradient in enthalpy flow rate we can relate the physical length of the duct to the electrical and thermodynamic parameters by

$$\frac{dX}{dH} = -\delta U_{\ell} \left[ \frac{S\beta_g}{v_g} + \frac{(1 - \beta_g)}{v_{\ell}} \right] = \frac{\left(\frac{t}{E_0}\right)^2}{\frac{E}{E_0} \left(1 - \frac{E}{E_0}\right)} \quad (\text{A } 11.2.35)$$

#### A 11.2.5 Steam Bottom Cycle

The steam bottom cycle is calculated in somewhat less detail than the MHD top cycle. The procedure amounts to assigning portions of the thermal energy rejected from the top cycle to

- Steam generation and superheat (if used)
- Steam reheat
- Combustion air preheat
- Rejected heat.

Internal reversibility factors are assigned to the bottom cycle for each component heat source used directly in the steam cycle, and a stack loss efficiency factor is assigned to the combustor-primary heat exchanger combination. The overall thermodynamic efficiency is calculated for each heat source component from the appropriate temperature profile according to the equation

$$\eta_{th} = 1 - T_0 \ln \left( \frac{T_1}{T_2} \right) / (T_1 - T_2) \quad (\text{A } 11.2.36)$$

Here  $T_0$  is the steam cycle sink temperature, and  $T_1$  and  $T_2$  are the upper and lower values along the top cycle gas temperature profile.

These components are combined to calculate overall bottom cycle power and efficiency and combined-cycle power and efficiency.

#### A 11.2.6 Input Data for Numerical Calculations

The following information is read as input into the program.

- Cycle configuration
  - The number of compressor stages each with a pre(inter) cooler ahead (up to four stages)
  - The number of regenerative heat exchangers (as either zero or one)
- Fluid properties
  - Gas molecular weight
  - Gas specific heat at constant pressure
  - Liquid density
  - Liquid specific heat
  - Liquid electrical resistivity
- Computational options
  - Overall combined-cycle power output may be specified together with instructions to iterate on duct inlet cross-section size (constant aspect ratio maintained).
  - The duct efficiency estimate internal to the program may be overridden by input of a constant duct efficiency.
  - Thermodynamic calculations may be followed by an additional routine that estimates component costs.
- Component characteristics
  - Separator efficiency as a fraction of gas and fraction of liquid channeled into the gas stream. Ideally these fractions would be 1.0 and 0.0, respectively.

- Polytropic efficiency of the compressor and efficiency of the liquid-metal pump
  - Pressure loss, as  $\Delta P/P$ , through each of the heat exchangers, the separator, and the mixer. Piping pressure losses may be lumped into these.
  - The duct loading as a ratio of working voltage to open-circuit voltage
  - Initial slip ratio at the duct inlet
  - The temperature difference across the regenerator. This difference is constant since the heat exchanger is linear.
  - Reduction factors to be applied to duct inlet and exit values from Sutton's formula of ohmic end losses
- Magnitude parameters
    - The inlet duct width, in the direction of the magnetic field, and the duct thickness in the direction of the electric field. The former is allowed to vary along the length to maintain constant liquid velocity, while the latter is held constant.
    - Output power may be specified. If iteration to this power is desired, the inlet duct dimensions will change, in the same ratio, to the size required for that power.
    - Inlet liquid velocity
    - Either the open-circuit voltage or the magnetic flux may be specified.
  - Fluid conditions
    - Mixture temperature and pressure, and gas volumetric flow fraction, are specified at the outlet of the mixer.
    - Any one of the above parameters is specified at the duct outlet.



- Temperature is specified at the gas stream outlet of the steam generator (the first heat exchanger after expansion in the duct), and a common gas temperature is given at each of the compressor inlets.
- The steam cycle bottom temperature is specified.
- Bottom and overall cycle characteristics
  - Reversibility index for each source and use of thermal energy into the steam cycle
  - Stack efficiency
  - Fraction of thermal energy from the steam generator (between Points 6 and 7) assigned to steam reheat and the temperature limits over which this is taken.

#### A 11.2.7 Calculational Procedures

##### A 11.2.7.1 MHD Cycle

The starting point for cycle analysis is the mixer exit (Point 1 in Figure A 11.2.1). The changes in temperature and enthalpy are determined from the ohmic heating as the mixture approaches the duct inlet (Point 2).

Depending on which duct outlet parameter is specified, the change in value over the duct length is divided into increments and a numerical integration carried out to determine conditions along the duct, making use of Equations A 11.2.13, A 11.2.14, or A 11.2.15. At each mesh point the slip ratio and the efficiency are calculated for use in the numerical integration. Also, the duct length integral and the duct cross-sectional area are calculated.

Again, ohmic heating at the duct outlet is estimated to give conditions at the separator inlet. The enthalpy change from Point 1 to Point 4 is the work performed by the duct.

#01-11

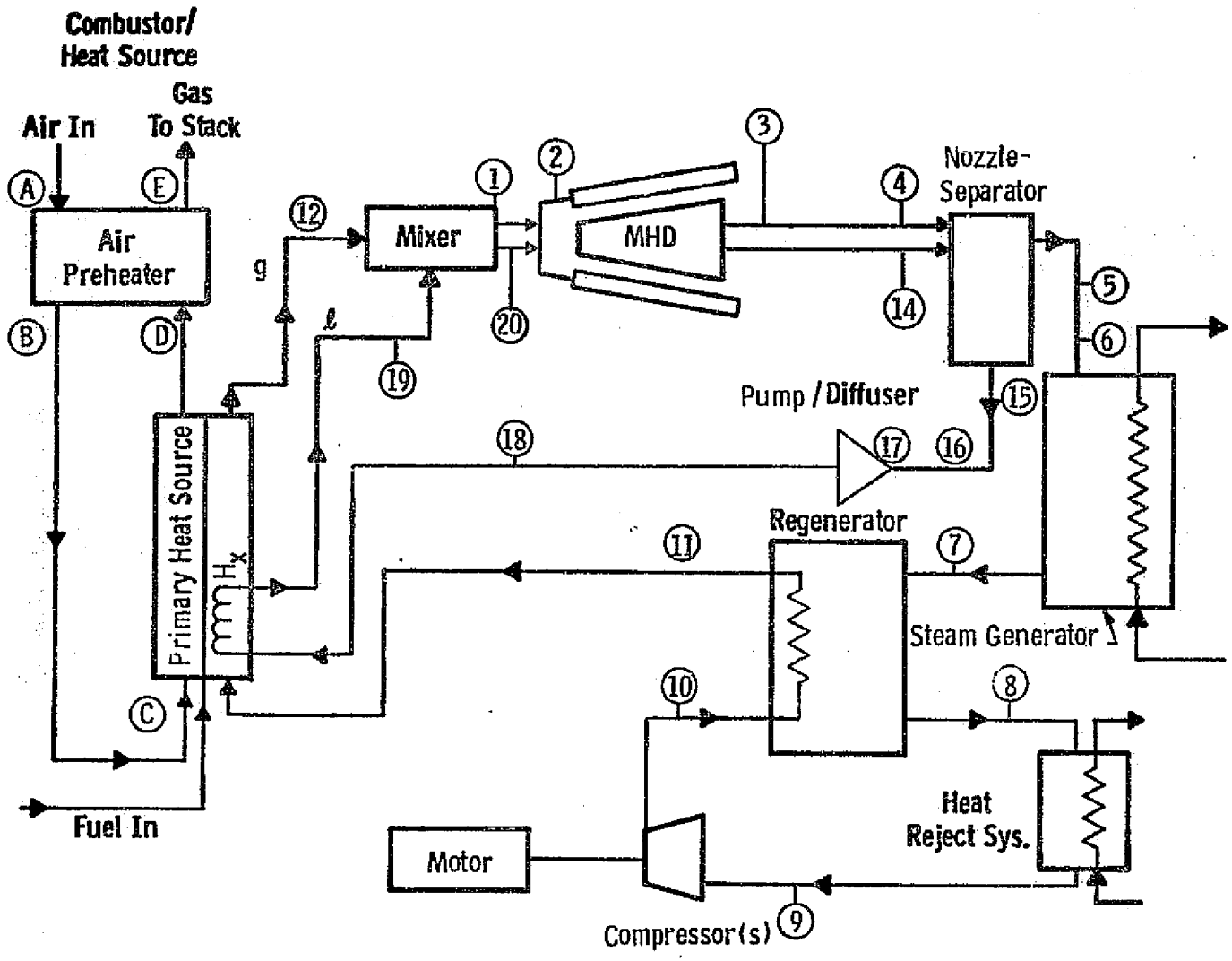


Fig. A 11. 2. 1 - Schematic arrangement of MHD cycle

The system component pressure losses allow the compressor pressure ratio to be calculated, which, combined with the compressor inlet temperature input, permits estimation of conditions at the exit of each compressor stage using the compression analogue of Equation A.2.12 with the input polytropic efficiency. Equal pressure ratios are assumed for each stage.

Temperatures are given at Points 7 and 9; thus the input approach temperature difference allows temperatures at Points 8 and 11 to be determined.

Proceeding along similar lines, we can calculate the outlet conditions for the liquid-metal pump. Here the pump work is increased by the reciprocal of pump efficiency, with the excess work appearing as thermal energy in the liquid metal.

To close the cycle it is only necessary to add enthalpy to one or both streams between Points 11 and 12 or Points 18 and 19 to return to mixer outlet enthalpy. The program calculates temperatures at the mixer inlet for various splits between heat addition to the two streams.

#### A 11.2.8 The Program

The listings of the program and subroutines are given in Subappendix AA 11.2.1. The input data are as follows:

ATWT	Molecular weight of the gas component
CP (1)	Specific heat at constant pressure of the gas component, Btu/lb-°F
DENS	Density of the liquid, lb/ft <sup>3</sup>
CP(2)	Specific heat of the liquid, Btu/lb-°F
RESIS	Resistivity of the liquid, ohm-cm
VEL	Liquid-metal duct velocity, ft/s
WIDE(1)	Duct inlet width in magnetic field direction, ft
THICK	Duct thickness in the electric field direction, dt
VLØ	Duct open-circuit potential, V

FLØ	Duct magnetic flux, T
VLØAD	Duct loading, V/V
SL	Duct inlet slip ratio
FACT(1), FACT(2)	Reduction factors for Sutton's formula for inlet
EFC	Compressor polytropic efficiency
EFP	Pump efficiency
PI(1,1)	Duct inlet pressure, psi
TF(1,1)	Duct inlet temperature, °F
F(1,1)	Duct inlet void fraction
PIØUT	Duct exit pressure, psi
TFØUT	Duct exit temperature, °F
FØUT	Duct exit void fraction
TF(1,9)	Compressor inlet temperature, °F
TSINK	Steam cycle sink temperature, °F
NCØMP	Number of compressor stages - 1, 2, 3, or 4
NREG	Number of regenerator stages - 0 or 1
THETA	Regenerator approach temperature difference, °F
TGEN	Gas stream temperature leaving steam generator at Point 7, °F
TFFD	Lowest temperature usable for feedwater heat, °F
FSEP(1), FSEP(2)	Fraction of gas and liquid entering the gas
DELP(4), DELP(6), DELP(7)	Fractional pressure losses in the gas stream through the separator, steam generator, and regenerator hot side, respectively
DELP(28), DELP(38), DELP(48), DELP(58)	Fractional pressure losses through the compressor precoolers
DELP(10), DELP(11), DELP(12)	Fractional gas stream pressure losses through the regenerator cold side, source heat exchanger, and mixer, respectively

DELP(14) Fractional pressure loss in liquid-metal stream  
 through the separator  
 DELP(16) Pressure loss through any additional components  
 in liquid-metal stream between the separator  
 and pump  
 DELP(18), } Fractional pressure loss in liquid stream  
 DELP(19) } through source heat exchanger and mixer  
 ETAS(6) } Internal reversibility index of the steam cycle  
 applicable to thermal energy fed from the steam  
 generator  
 ETAS(28), } Internal reversibility indices of steam cycle for  
 ETAS(38), } thermal energy entering the cycle through the  
 ETAS(48), } compressor precoolers, combined with the fraction  
 ETAS(58) } of the heat used in this way  
 EFSTCK Stack efficiency  
 ETAPRE(28), } Fraction of the thermal energy from the compressor  
 ETAPRE(38), } precoolers used for combustion air preheat. Note  
 ETAPRE(48), }  $ETAPRE(N) + ETAS(N) < 1$   
 ETAPRE(58) }  
 REHEAT Fraction of thermal energy from the steam  
 generator (between Points 6 and 7) used for  
 steam reheat  
 TFUP, } Upper and lower limits (bottom cycle hole) over  
 TFDOWN } which thermal energy is used as steam reheat  
 EFRHH Internal reversibility index for thermal  
 energy used as steam reheat  
 NCGST Index for cost routine option 1 if cost routine  
 is to be used, 0 otherwise  
 NITER Index for option to iterate on power. 1 if  
 iteration is desired, 0 otherwise

**POWER** Overall combined-cycle power for which component sizes will be chosen, MW

**EEDUCT** MHD duct efficiency to be taken as fixed if nonzero value is entered.

A sample output is given in Subappendix AA 11.2.2.

A 11.2.9 Nomenclature

B	Magnetic flux density
$C_D$	Interphase drag coefficient
$C_{\ell}$	Specific heat of liquid metal
$C_{PG}$	Specific heat of gas at constant pressure
$C_{VG}$	Specific heat of gas at constant volume
$D_g$	Interphase drag force acting on gas component per unit volume gas
$D_{\ell}$	Interphase drag force acting on liquid component per unit volume gas
E	Working potential across duct thickness
$E_0$	Open-circuit potential across duct thickness
$\epsilon_c$	Volumetric energy conversion rate
$\epsilon_d$	Volumetric energy degradation rate
$f_{\ell}$	Electromagnetic body force acting on liquid metal
g	Gravitational constant
H	Mixture enthalpy
j	Current density
P	Pressure
p	Power
R	Gas law constant
$R_E$	Effective and electrical resistance
$R_g$	Gas momentum velocity gradient
$R_{\ell}$	Liquid momentum velocity gradient
S	Slip ratio, $U_g/U_{\ell}$
T	Temperature
t	Duct thickness normal to the magnetic field
$U_g$	Gas-phase velocity
$U_{\ell}$	Liquid velocity

$v_g$	Gas specific volume
$v_l$	Liquid specific volume
$W$	Width of duct parallel to magnetic field
$w_g$	Fraction of total mass flow as gas
$w_l$	Fraction of total mass flow as liquid
$X$	Coordinate along duct length direction
$\beta_g$	Fraction of cross-sectional area occupied by gas
$\beta_l$	Fraction of cross-sectional area occupied by liquid
$\gamma$	Weighted mean specific heat ratio of the mixture
$\eta_i$	Isentropic expansion efficiency
$\eta_p$	Polytropic expansion efficiency
$\eta_s$	Slip efficiency factor
$\eta_{th}$	Ideal thermodynamic efficiency
$\kappa$	Electrical conductivity of the pure liquid metal
$\rho_g$	Gas density
$\rho_l$	Liquid density
$\delta$	Electrical resistivity of the mixture
$\phi_g$	Fraction of total volume flow as gas
$\phi_l$	Fraction of total volume flow as liquid



Subappendix AA 11.2.1

COMPUTER PROGRAM LISTING

WRUN,N/RNPT MHDTOP,09E76ECASAO,GRIMBLE,1,25

RSYM PRINTS,PR7

QASG,A GRIMBLE\*MHD.

WFOR,IS MHD\*MHDTOP  
FOR DT2B-U6/10/75-11:05:08 (.0)

MAIN PROGRAM

STORAGE USED: CODE(1) 002145; DATA(0) 002656; BLANK COMMON(2) 010165

EXTERNAL REFERENCES (BLOCK, NAME)

0003	VALUE
0004	SHUNT.
0005	FIXP
0006	FIXV
0007	FIXT
0010	SKIP
0011	SEPER
0012	THROT
0013	SINK
0014	REPORT
0015	HEAT
0016	COST
0017	NINTRS
0020	NLOCBS
0021	NSCR
0022	NNRFUS
0023	NNWFUS
0024	ALOG
0025	XPRR
0026	SQRT
0027	NSTOP*
0030	NBPOS

11-112

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000	000267	1F	0000	000421	10E	0001	000006	100L	0001	001121	1005G	0001	001234	1042G
0001	001240	1046G	0001	001273	1055G	0001	000322	107L	0001	001322	1070G	0001	000521	108L
0000	000434	11F	0001	000530	112L	0000	002313	1120D	0000	002317	1124D	0000	002323	1127D
0000	002327	1135D	0000	002334	1144D	0000	002341	1150D	0000	002346	1154D	0001	001541	1157G
0001	000536	116L	0000	002352	1161D	0000	002366	1174D	0000	002372	1177D	0000	000447	118F
0000	002376	1202D	0000	002402	1205D	0000	002406	1210D	0000	002412	1216D	0001	001627	1223G
0000	002416	1232D	0000	002423	1236D	0000	002430	1242D	0000	002435	1247D	0000	002442	1253D
0000	002446	1256D	0000	002452	1261D	0000	002456	1264D	0000	002462	1270D	0000	002442	1273D
0000	002474	1276D	0000	002501	1301D	0000	002506	1304D	0000	002513	1310D	0000	002524	1317D
0000	002531	1324D	0000	002536	1327D	0000	002543	1332D	0000	002550	1335D	0000	002554	1337D
0001	002062	1343G	0000	002560	1356D	0000	002567	1366D	0001	001102	137L	0001	001126	137L
0000	001644	143D	0001	001174	146L	0000	000457	15F	0000	000467	16F	0000	000477	17F
0000	000506	18F	0000	000516	19F	0001	002141	199L	0000	000276	20F	0001	000260	202L
0001	000343	203L	0001	000350	204L	0001	001516	208L	0001	001527	212L	0000	000653	214D
0001	001533	214L	0000	001665	223D	0001	001620	228L	0000	001474	230D	0000	001704	236D
0000	001714	244D	0000	000526	25F	0000	001723	251D	0000	000612	254F	0000	001732	256D
0000	001740	262D	0000	000770	266F	0000	001751	271F	0000	001060	271F	0000	001757	275D
0000	001217	283F	0000	001430	290F	0000	001473	291F	0000	001561	295F	0000	000305	30F
0000	001767	303D	0000	000344	307F	0000	000376	308F	0000	001777	311D	0000	002006	36D
0000	002016	324D	0000	002030	334D	0000	002041	334D	0000	002051	351D	0000	000431	355D
0000	002063	362D	0000	002065	364D	0000	002071	366D	0000	002077	372D	0000	002106	414D
0000	001243	383F	0000	000314	40F	0000	002114	403D	0000	002122	407D	0000	002131	444D
0000	002137	420D	0000	002146	425D	0000	002152	427D	0000	002162	435D	0000	002172	443D
0000	002201	450D	0000	002205	452D	0000	000650	454F	0000	002215	460D	0000	002227	470D
0000	002240	477D	0000	001267	483F	0000	001564	495F	0000	000324	50F	0000	002250	545D
0000	000551	52F	0000	002257	522D	0000	002266	527D	0000	002274	533D	0000	000801	545F
0000	002304	541D	0001	000366	567G	0001	000367	572G	0000	000334	60F	0000	000673	60F
0001	000402	602G	0000	000704	61F	0001	000424	613G	0000	000717	62F	0000	000721	64F
0000	000740	65F	0000	000757	66F	0000	001001	68F	0000	001020	69F	0000	000354	70F
0000	001034	70F	0001	000654	703G	0001	000661	706G	0000	001052	71F	0000	001066	72F
0001	000706	721G	0001	000750	731G	0001	000757	734G	0000	001107	74F	0001	001031	747G

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00343 121 READ(5,98) REHEAT,TFUP,TFDOWN,EFRMT
00351 122 READ(5,295) NCOST,NITER,POWER,EFDUCT
00357 123 VOLT=VLD
00360 124 FLUX=FLU
00361 125 ETAS(8)=ETAS(28)
00362 126 WRITE(6,71)
00364 127 WRITE(6,80)
00366 128 WRITE(6,88) ATWT,CP(1)
00372 129 WRITE(6,89) DENS,CP(2),RESIS
00377 130 WRITE(6,72) FACT(1),FACT(2)
00403 131 WRITE(6,74) EFC,EFP
00407 132 WRITE(6,81) TF(1,1),PI(1,1),F(1,1)
00414 133 WRITE(6,87) FSEP(1),FSEP(2)
00420 134 WRITE(6,91) TF(1,9),EFC,TSINK
00425 135 WRITE(6,92)
00427 136 WRITE(6,93) DELP(4),DELP(6),DELP(7),DELP(24)
00435 137 WRITE(6,83) DELP(28),DELP(38),DELP(48),DELP(58)
00443 138 WRITE(6,94) DELP(10),DELP(11),DELP(12)
00450 139 WRITE(6,96)
00452 140 WRITE(6,77) DELP(14),DELP(16),DELP(18),DELP(19)
00460 141 WRITE(6,283) ETAS(6),ETAS(26),ETAS(38),ETAS(50),ETAS(5)
00470 142 WRITE(6,463) EFSTCK,ETAPRE(28),ETAPRE(38),ETAPRE(58)
00477 143 WRITE(6,495) REHEAT,TFUP,TFDOWN,EFRMT
00505 144 TAB=459.69
00506 145 TRFD=TFD+TAB
00507 146 TREF=TSINK+TAB
00510 147 TRUP = TFUP + TAB + 0.0001
00511 148 TRDOWN = TFDOWN + TAB
00512 149 ETATHR = 1.-TREF * LOG(TRUP/TRDOWN)/(TRUP-TRDOWN)
00513 150 KOUNT=D
202 KOUNT=KOUNT+1
00514 151 WRITE(6,98) VEL,WIDE(1),THICK
00515 152 WRITE(6,290) FLUX,VLOAD,SL
00524 153 WRITE(6,291) VOLT,WVOLT
00533 154 WRITE(6,25) HCOMP,NREG,THETA,TGEN
00541 155 WRITE(6,55) POWER,NITER,NCOST
00546 156 IF(KOUNT-GE,2) GO TO 100
00550 157 RHEAT=RHEAT/ATWT
00551 158 RWORK=RHEAT/.001248
00552 159 PIF=3.14159
107 PIF(3)=PIOUT
00553 160 TF(1,3)=TFOUT
00554 161 F(1,3)=FOUT
00555 162 IF(VLD.LE,0.001) GO TO 203
00556 164 FLUX=VOLT*(VEL*THICK*.0929)
00560 165 GO TO 204
203 VOLT=VEL*THICK*FLUX*.0929
00561 166
204 CONTINUE
00562 167
00563 168 WVOLT=VOLT*VLOAD
00564 169 COND=30.48/RESIS
00565 170 DO 104 L=1,3
00566 171 DO 102 J=2,3
00571 172 TF(J,L)=TF(1,L)
00574 173 P(1,J,L)=P(1,L)
00575 174 F(2,J,L)=F(1,L)
102 F(3,J,L)=F(1,L)
00577 175 DO 104 J=1,3
00600 176 P(J,L)=P(J,L)+144
00601 177 TR(J,L)=TF(J,L)+TAB
00604 178
00605 179
104 CONTINUE
00611 180 V(1,1)=RWORK*YR(1,1)/PF(1,1)
00612 181 DO 105 L=1,90
00615 182 V(2,L)=1./DENS
105 VRAT=V(1,1)/V(2,1)
00617 183 F(2,1)=1-F(1,1)
00620 184 F(3,1)=1
00621 185 WT(1,1)=1/(1+VRAT*F(2,1)/F(1,1))
00622 186 WT(2,1)=1-WT(1,1)
00623 187 WT(3,1)=1
00624 188 V(3,1)=(V(1,1)*WT(1,1)+V(2,1)*WT(2,1))/WT(3,1)
00626 189 CV=CP(1)-RHEAT
00627 190 CAPP=CP(1)*WT(1,1)+CP(2)*WT(2,1)
00630 191 Y=RHEAT*WT(1,1)/CAPP
00631 192 FLHD=WIDE(1)*THICK*VEL/V(3,1)
00632 193 CALL VALUE(1)
00633 194 CALL SHUNT(1,WIDE(1),FACT(1))
00633 196

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00639 197* CALL VALUE(2)
00635 198* IF(PF(1,3)--001)108,108,106
00640 199* 106 CALL FIXP(2)
00641 200* GO TO 116
00642 201* 108 IF(F(1,3)--001)112,112,110
00645 202* 110 CALL FIXV(2)
00646 203* GO TO 116
00647 204* 112 IF(TR(1,3)--1)116,116,114
00652 205* 114 CALL FIXT(2)
00653 206* 116 CONTINUE
00654 207* CALL VALUE(3)
00655 208* CALL SHUNT(3, WIDE(2),FACT(2))
00656 209* CALL VALUE(4)
00657 210* ETAN=TR(1,1)-TR(1,4)/(TR(1,1)*(1-(PF(1,4)/PF(1,1)))*( Y))
00660 211* CALL SKIP(4,14)
00661 212* CALL VALUE(14)
00662 213* CALL SEPER(4,FSEP(1),FSEP(2))
00663 214* CALL VALUE(15)
00664 215* CALL VALUE(15)
00665 216* DP=-DELPI(4)
00666 217* CALL THROT(5,6,DP)
00667 218* CALL VALUE(6)
00670 219* TR(1,7)=TGEN+TAB
00671 220* DP=-DELPI(6)
00672 221* CALL SINK(6,7,TR(1,7),DP)
00673 222* CALL VALUE(7)
00674 223* DP=-DELPI(7)
00675 224* CALL THROT(7,8,DP)
00676 225* HOUT=H(3,6)-H(3,7)
00677 226* ENER=ETATH(6)*ETAS(6)*(H(3,6)-H(3,7))
00700 227* THIN=TF(1,9)+459.69
00701 228* CALL SKIP(1,13)
00702 229* DO 120 L=12,10,-1
00705 230* DO 120 J=1,3,1
00710 231* WT(J,L)=WTIJ(8)
00711 232* PF(J,L)=PF(J,L+1)/(1-DELPI(L))
00712 233* 120 CONTINUE
00715 234* CALL SKIP(8,28)
00716 235* CALL VALUE(28)
00717 236* PRAT=PF(1,10)/PF(1,8)
007 237* DO 121 N=1,NCOMP,1
00723 238* LC=(N+1)*10+8
00724 239* 121 PRAT*PRAT/(1-DELPI(LC))
00726 240* PRAT=PRAT*(1-DELPI(LC))
00727 241* Y=WT(1,8)*RHEAT/WT(1,8)*CP(1)+WT(2,8)*CP(2)
00730 242* DO 134 N=1,NCOMP,1
00733 243* DO 130 KAP=8,10,1
00736 244* 130 LCOMP(KAP)=(N+1)*10+KAP
00740 245* LCB=LCOMP(8)
00741 246* LCP=LCOMP(9)
00742 247* LC10=LCOMP(10)
00743 248* DP=-DELPI(LCB)
00744 249* CALL SINK(LCOMP(8),LCOMP(9),THIN,DP)
00745 250* CALL VALUE(LCOMP(9))
00746 251* DO 132 J=1,3,1
00751 252* TR(J,LC10)=TR(1,LC9)*PRAT*(Y/EFCL)
00752 253* PF(J,LC10)=PF(1,LC9)*PRAT
00753 254* 132 WT(J,LC10)=WT(J,LC9)
00755 255* CALL VALUE(LCOMP(10))
00756 256* LT=LCOMP(10)+8
00757 257* CALL SKIP(LCOMP(10),LT)
00760 258* CALL VALUE(LT)
00761 259* 134 CONTINUE
00763 260* CALL SKIP(LCOMP(10),10)
00764 261* CALL VALUE(10)
00765 262* NOT=NRG
00766 263* IF(TR(1,10)+THETA,GE,TR(1,7))NOT=0
00770 264* 135 CALL SKIP(10,11)
00773 265* GO TO 139
00774 266* 137 DO 136 J=1,3,1
00775 267* TR(J,8)=TR(1,10)+THETA
00776 268* TR(J,7)=TGEN+TAB
00777 269* 136 TR(J,11)=TR(J,7)-THETA
00778 270* DO 138 L=7,8,1
00779 271* CALL VALUE(L)
00780 272*
00807 00807
00805 00805
00804 00804
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01011 273* CALL VALUE(11)
01012 274* CALL VALUE(8)
01013 275* IF(ITFFD=.0001)146,146,142
01016 276* 142 CALL SKIP(8,26)
01017 277* CALL VALUE(26)
01020 278* DP=-DELP(26)
01021 279* CALL SINK(26,8,TRFD,OP)
01022 280* CALL VALUE(8)
01023 281* ENER=ENER+ETATH(26)*ETAS(26)*(H(3,26)-H(3,8))
01024 282* HOUT=HOUT+H(3,26)-H(3,8)
01025 283* CALL SKIP(8,28)
01026 284* CALL VALUE(28)
01027 285* 144 CONTINUE
01030 286* DP=-DELP(28)
01031 287* CALL SINK(8,29,THIN,DP)
01032 288* DP=-DELP(14)
01033 289* CALL THROT(15,16,DP)
01034 290* CALL VALUE(16)
01036 291* DP=-DELP(16)
01037 292* CALL THROT(16,17,DP)
01040 293* CALL SKIP(1,20)
01041 294* DO 122 J=1,3,1
01044 295* WT(J,18)=WT(J,17)
01045 296* DO 122 L=19,18,-1
01050 297* PF(J,L)=PF(J,L+1)/(1-DELP(L))
01051 298* CONTINUE
01054 300* DO 123 J=1,3,1
01057 301* 123 TR(J,18)=TR(J,17)+(PF(2,18)-PF(2,17))*V(3,17)*.001284*(1/EFPP-1)
01057 302* 1 WT(3,17)/(CP(1)*WT(1,17)+CP(2)*WT(2,17))
01061 303* CALL VALUE(18)
01062 304* HIN=H(3,1)+H(3,18)-H(3,11)
01063 305* WORK=H(3,1)+H(3,17)-H(3,18)-H(3,17)
01064 306* ETAPRE(18)=ETAPRE(28)
01065 307* PREHT = 0
01066 308* HSTH=HOUT
01067 309* DO 226 N=1,NCOMP,1
01072 310* LCB=(N+1)*10*8
01073 311* TF(N,EQ-1)LCB=8
01075 312* LC9=(N+1)*10*9
01076 313* LC10=(N+2)*10
01077 314* WORK=WORK-(H(3,LC10)-H(3,LC9))
01100 315* ENER=ENER+ETATH(LCB)*(H(3,LCB)-H(3,LC9))*ETAS(LCB)
01101 316* PREHT=PREHT+(H(3,LCB)-H(3,LC9))*ETAPRE(LCB)
01102 317* HSTH=HSTH+(H(3,LCB)-H(3,LC9))*ETAS(LCB)/(ETAS(LCB)+.0001)
01103 318* 226 HOUT=HOUT+(H(3,LCB)-H(3,LC9))
01105 319* HREHT = HREHT + HSTH / (1.0+HREHT)
01106 320* ENER = ENER + HREHT * ETATHR * EFRHT
01107 321* ETATH=ENER/(HSTH+HREHT)
01110 322* ENER=ENER+WORK
01111 323* POW=WIDE(1)*THICK*VEL*3.6*ENER/(V(3,21)*3412)
01112 324* RAT=SQRT(POWER/POW)
01113 325* WIDE(1)=WIDE(1)*RAT
01114 326* THICK=THICK/RAT
01115 327* ITABS(ITER)*(RAT-1)-.0001)127,127,107
01120 328* 127 WRITE(6,70)
01122 329* NREG=HOT
01123 330* POWER=POW
01124 331* WRITE(6,11)
01126 332* CALL REPORT(1)
01127 333* WRITE(6,12)
01131 334* CALL REPORT(2)
01132 335* IF(FOUT=.001)208,208,206
01135 336* 206 WRITE(6,82)F1OUT
01140 337* GO TO 216
01141 338* 208 IF(FOUT=.001)212,212,210
01144 339* 210 WRITE(6,86)FOUT
01147 340* GO TO 216
01150 341* 212 WRITE(6,84)TFOUT
01153 342* 216 CONTINUE
01154 343* WRITE(6,52)
01156 344* DO 124 I=1,40,1
01161 345* 124 WRITE(6,62)DIST(I),*IDTH(I),TEMP(I),PRES(I),BETA(I),FGAS(I),
01161 346* 1 SLIP(I),EFF(I)
01174 347* WRITE(6,3)
01176 348* CALL REPORT(3)
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01177	149*	WRITE(6,4)			001553
01201	150*	CALL REPORT(4)			
01202	151*	WRITE(6,5)			001561
01205	152*	CALL REPORT(6)			
01207	153*	WRITE(6,307)			001567
01212	154*	IF(NREG.GE.1)WRITE(6,7)			
01213	155*	CALL REPORT(7)			
01216	156*	IF(TFFD=.001)226,228,227			001605
01220	157*	227 WRITE(6,308)			001611
01221	158*	CALL REPORT(26)			
01222	159*	228 CONTINUE			001620
01223	160*	DO 126 N=1,NCOMP,1			001620
01225	161*	LC8=(N+1)*10+8			001632
01226	162*	IF(N.EQ.1)LC8=8			001635
01230	163*	LC9=(N+1)*10+9			001642
01231	164*	LC10=(N+2)*10			001645
01232	165*	WRITE(6,8)N			001651
01235	166*	CALL REPORT(LC8)			
01236	167*	WRITE(6,9)N			001657
01241	168*	CALL REPORT(LC9)			
01242	169*	WRITE(6,10)N			001665
01245	170*	CALL REPORT(LC10)			
01246	171*	ETAC=TR(1,LC9)	)=(PRAT*Y-1)/(TR(1,LC10	)-TR(1,LC9))	001673
01247	172*	126 WRITE(6,6)ETAC			001713
01253	173*	WRITE(6,11)			001721
01255	174*	CALL REPORT(11)			
01256	175*	WRITE(6,15)			001727
01260	176*	CALL REPORT(16)			
01261	177*	WRITE(6,17)			001735
01263	178*	CALL REPORT(17)			
01264	179*	WRITE(6,18)			001743
01266	180*	CALL REPORT(18)			
01267	181*	ETAHHD=WORK/HIN			001751
01270	182*	WRITE(6,61)HIN			001754
01273	183*	WRITE(6,64)HOUT			
01276	184*	WRITE(6,75)WORK			
01301	185*	WRITE(6,54)ETAH			
01304	186*	WRITE(6,66)ETAHHD			
01307	187*	ETATOT = ENER * EFSTCK / (HIN + HREHT - EFSTCK * PREHT)			
01310	188*	WRITE(6,38)ETATH(6),ETATH(28),ETATH(38),ETATH(48),ETATH(58)			002007
01317	189*	WRITE(6,264)ETAST			
01322	190*	ETA1=ENER/(HIN+HREHT-PREHT)			002024
01323	191*	ETA2=ENER/(HIN+HREHT)			002027
01324	192*	WRITE(6,354)ETA1			
01327	193*	WRITE(6,454)ETA2			
01332	194*	WRITE(6,254)ETATOT			
01335	195*	WRITE(6,68)			
01337	196*	WRITE(6,67)			
01341	197*	DH=H(3,1)-H(3,18)			002062
01342	198*	DO 128 I=9,10,1			002064
01345	199*	DELH(11)=DH*1/10,			002071
01346	400*	DELH(18)=DH-DELH(11)			002073
01347	401*	DP=-DELP(11)			002075
01350	402*	CALL HEAT(11,12,DELH(11),DP)			002102
01351	403*	CALL VALUE(12)			002104
01352	404*	DP=-DELP(18)			002113
01353	405*	CALL HEAT(18,19,DELH(18),DP)			002115
01354	406*	CALL VALUE(19)			002120
01355	407*	PROP=1/10,			002126
01356	408*	128 WRITE(6,62)PROP,TF(1,12),TF(2,19)			002134
01364	409*	IF(NCOST.GE.1)CALL COST			
01366	410*	WRITE(6,271)			
01370	411*	GO TO 202			
01371	412*	199 CONTINUE			002141
01372	413*	STOP			002141
01373	414*	END			002144

END OF COMPILATION: NO DIAGNOSTICS.

\$FIN



SUBROUTINE EFIZNC ENTRY POINT 000375

STORAGE USED: COUL(1) 000408; DATA(1) 000475; BLANK COMMON(2) 010165

LABELLED REFERENCES (BLOCK, NAME)

0003 LAP  
0004 SUR1

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000024	340L	0001	000176	J50L	0001	000345	370L	0000	000040	ΔZF	0000	R	000022	AA
0002	000180		0001	000023	BB	0002	R	010011	BLTA	0000	R	000036	ROT	0000	CC
0002	000197	COND	0002	000715	CP	0002	R	007721	CT	0000	R	006535	DELH	0002	DEL
0002	000487	DELTA	0001	000021	CL5DEL	0002	R	007545	CYST	0000	R	000037	DPC	0000	DS
0000	000048	EF11	0001	010104	EF0BLT	0002	R	007627	E1F	0000	R	000004	EFF1	0002	ETA5
0002	006267	E1ATH	0002	002208	F1V	0000	R	000021	FAVE	0002	R	007401	FGAS	0002	FLN
0002	006250	F1LHO	0002	005632	FLV	0002	R	003124	H	0000	R	000013	HDEL	0000	J
0000	000000	F1LHO	0001	000012	KAP	0000	R	000027	KRD1	0000	R	000030	KHD2	0000	I
0000	000000	F1L	0002	000726	KCOLT	0000	R	000015	PDEL	0002	R	001452	PF	0000	R
0002	007153	TR5	0000	000015	KRES15	0002	R	010073	KG	0002	R	007723	KHEAT	0002	H
0002	003542	SL	0002	010163	SL	0000	R	000032	SLAVE	0002	R	007722	SLIP	0002	H
0000	000025	SLK	0000	000035	TAVE	0002	R	007235	TEMP	0002	R	000000	TF	0002	H
0000	000000	TRK	0002	000416	TR	0002	R	007725	TRF	0002	R	002070	V	0000	R
0002	010160	TRF	0002	001156	VLOAB	0002	R	007317	VCL	0000	R	000017	VOLDEL	0002	R
0002	007711	WIDE	0002	007463	WIDTH	0002	R	010162	WKAT	0002	R	004576	WT	0002	R

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SUBROUTINE EFIZNC(1,1)  
COMMON (J,90),P(1,90),P1(3,90),PF(3,90),Y(3,90),F(3,90),  
R(3,90),S(3,90),B(3,90),WT(3,90),FLM(3,90),FLV(3,90),FLHO,  
MLP(10),DELL(90),DELH(10),E1ATH(10),E1AS(10),  
RES(50),TEMP(50),VCL(50),FGAS(50),WIDTH(50),DIST(50),  
PF(50),WID(14),C(14),CY,RRDRK,KHEAT,Y,1REF,PCOEF,  
SLIP(50),ETA(50),KG(50),THICK,VLOAB,COND,VCL,VOLT,WKAT,SL,  
EFOC1  
DIMENS(10),PC(4),EFF(4)  
PCOAT(1,9,0,14,4)  
PCOEF= 300.3040 /V(2,L)  
K=1  
IF (EFOC1\*QLE+D\*0001) K=1  
PC(1)=PCOEF  
PC(2)=PCOEF+1.4  
J=1  
KAP=0  
J=J+1  
340 J=J+1  
HDEL=(TEMP(I)-TEMP(I-1))\*WRAT\*(C(1)+(1-WKAT)\*C(2))  
+1\*WKAT\*(PRES(I)-PRES(I-1))+RES(I-1)\*Y(2,L)\*0.01284  
1 RES(5)=LAP(3,0\*0.1\*1\*1)/C(0\*0)  
0.5\*(I)=D(I)-1\*((THICK/VCL)+2)\*RES(5)\*VCL\*HDEL\*  
1 (SLIP(I)-1)\*SLTA(I-1)/VCL(I-1)+(I-ETA(I-1))/V(2,L)/V  
2 (VLOAB\*(I-VLOAB)\*0.00948)  
PC(1)=(PRES(I)-PRES(I-1))/2.5  
VAVE=(VCL(I)+VCL(I-1))/2.  
VCLDEL=VCL(I)-VCL(I-1)  
0.5DEL=(VCL(I)-0.5\*(I-1)  
FAVE=(FGAS(I)+FGAS(I-1))/2  
AA=FAVE/2.1\*FAVE  
HB=VCLL/(VCL+VCL\*PC(J)\*D15DEL)  
CC=VOLDEL/(FAVE\*VAVE\*PC(J)\*D15DEL)  
SL=SLIP(I-1)  
SL2=SL\*1.2  
0.011=0.5\*WT(1)\*((SL\*AA/SL1)+3)\*(BB+CC\*SL1\*SL1)-SL1  
350 0.021=0.5\*WT(1)\*((SL2\*AA/SL2)+3)\*(BB+CC\*SL2\*SL2)-SL2  
DS=URD\*SL2-SL1/10RD2-URD1  
0.01=URD2  
SL1=SL2

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00141	40*	SL2=SL2+DS	000222
00142	41*	IF (ABS(DS/SL2).GT.0.00005)GO TO 350	000224
00143	42*	SLIP(1)=SL2	000231
00145	43*	SLAVE=(SLIP(1)+SLIP(I-1))/2	000232
00146	44*	TAVE=(TEMP(11)+TEMP(I-1))/2	000235
00147	45*	PAVE=(PRES(11)+PRES(I-1))*32.16/2	000236
00150	46*	RG(I)=VLE*VEL*SLAVE*(SLAVE*VOLUME)/DISDEL	000241
00151	47*	BETA(1)=FGAS(11)/(FGAS(11)+SLIP(1)+(1-FGAS(1)))	000246
00152	48*	DELTA(1)=RG(I)+PDEL/DISDEL	000255
00153	49*	BOT=(RG(I)+PDEL/DISDEL)*BETA(1)*SLIP(1)	000264
00154	50*	EFF(I,J)=TOP/BOT+PDEL*(1-BETA(1))/DISDEL	000272
00155	51*	IF (IAP*K-1)370,340,344	000276
00160	52*	344 DPC=(PC(J)-PC(J-1))*(EFF(I,J)-EFDUCT)/(EFF(I,J)-EFF(I,J-1))	000306
00161	53*	PC(1)=PC(2)	000313
00162	54*	PC(2)=PC(2)+DPC	000323
00163	55*	EFF(I,I)=EFF(I(2))	000325
00164	56*	J=J-1	000327
00165	57*	IF (ABS(EFF(I,J)-EFDUCT)/EFDUCT).GT.0.001)GO TO 340	000331
00167	58*	EFF(I,I)=EFF(I(1))	000334
00170	59*	IF (I.EQ.1)EFF(I)=EFDUCT	000345
00172	60*	WIDTH(1)=WIDE(1)*(1-BETA(1))/(1-BETA(1))	000346
00173	61*	RLTURN	000353
00174	62*	END	000362
			000405

END OF COMPILATION:            NO DIAGNOSTICS.

SUBROUTINE EQUATE ENTRY POINT 000061

STORAGE USED: CODE(1) 000071; DATA(0) 000016; BLANK COMMON(2) 010164

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	0003, 2	1050	0001	000036	1216	0002	004160	B	0002	010011	BETA	0002	010157	COND			
0002	007715	CP	0002	007721	CV	0002	006535	DELH	0002	006251	DELP	0002	006403	DELT			
0002	007515	D151	0000	000003	DTMS	0002	007627	EFF	0002	007021	ETAS	0002	006667	ETATH			
0002	K	002500	F	0002	K	007401	FGAS	0002	005214	FLH	0002	006250	FLMO	0002	005632	FLV	
0002	003124	H	0000	1	000000	J	0002	007726	PCOEF	0002	K	001452	PF	0002	001034	PI	
0002	K	007153	PRES	0002	010073	RG	0002	007723	RHEAT	0002	007722	RWORK	0002	003542	S		
0002	010103	SL	0002	007727	SLIP	0002	K	007235	TAMP	0002	000000	TF	0002	010155	THICK		
0002	K	000416	TR	0002	007725	TREF	0002	K	002070	V	0002	010160	VEL	0002	010156	VLOAD	
0002	K	007317	VOL	0002	010161	VOLT	0002	K	007711	WIDE	0002	K	007463	WIDTH	0002	010162	WRAT
0002	K	004576	WT	0002	007724	Y											

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00101 1.      SUBROUTINE EQUATE(L)
00103 2.      COMMON IF(3,90),TR(3,90),PI(3,90),PF(3,90),V(3,90),F(3,90),
00103 3.      H(1,90),S(3,90),B(3,90),WT(3,90),FLH(3,90),FLV(3,90),FLMO,
00103 4.      DELP(90),DELT(90),DELH(90),ETATH(90),ETAS(90),
00103 5.      PRES(50),TEMP(50),VOL(50),FGAS(50),WIDTH(50),DIST(50),
00103 6.      LIP(50),WIDE(4),CP(4),CV,RWORK,RHEAT,T,TREF,PCOEF,
00103 7.      SLIP(50),DETA(50),RG(50),THICK,VLOAD,COND,VEL,VOL1,WRAT,SL
00103 8.      DO 804 J=1,3,1
00107 9.      TR(J,L+1)=TAMP(40)
00110 10.     804 PF(J,L+1)=PRES(40)
00112 11.     WIDE(2)=WIDTH(40)
00113 12.     F(1,L+1)=FGAS(40)
00114 13.     F(2,L+1)=F(40)
00115 14.     F(3,L+1)=1.
00116 15.     V(1,L+1)=VOL(40)
00117 16.     V(2,L+1)=V(2,L)
00120 17.     DO 806 J=1,3,1
00123 18.     806 H(J,L+1)=V(J,L)
00125 19.     V(3,L+1)=(V(1,L+1)*WT(1,L+1)+V(2,L+1)*WT(2,L+1))/WT(3,L+1)
00126 20.     RETURN
00127 21.     END

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END OF COMPILATION: NO DIAGNOSTICS.

SUBROUTINE FIX ENTRY POINT 000130

STORAGE USED: CODE(1) 000146; DATA(1) 000026; BLANK COMMON(2) 010164

EXTERNAL REFERENCES (BLOCK, NAME)

0003 EF1ZNC  
 0004 EQUATE  
 0005 XPRK

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000050	1216	0004	004160	B	0002	R	010011	BETA	0002	010157	COND	0002	007715	CW		
0002	007721	CV	0002	006535	DELH	0002	R	006251	DELP	0002	006403	DELT	0002	R	007545	DIST	
0000	R	000000	0000	000004	DYNS	0002	R	007627	EFF	0002	007021	ETAS	0002	006667	ETATH		
0002	R	002504	0002	R	007401	FGAS	0002	005214	FLM	0002	006250	FLMU	0002	005632	FLV		
0002	003124	H	0000	I	000002	I	0000	I	000001	L	0002	007726	PCOEF	0002	R	001452	PF
0002	001034	PI	0002	R	007553	PKES	0002	R	010073	RG	0002	007723	RHEAT	0002	R	007722	RWORK
0002	003542	S	0002	R	010163	SL	0002	R	007727	SLIP	0002	R	007235	TEMP	0002	000000	TE
0002	010155	THICK	0002	R	000416	TR	0002	007725	TREF	0002	R	002070	V	0002	010160	VEL	
0002	R	010156	0002	R	007317	VOL	0002	010161	VOLT	0002	R	000003	VKAT	0002	R	007711	WIDE
0002	R	007463	0002	R	010162	WRAT	0002	R	009576	WT	0002	R	007724	Y			

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00101 1* SUBROUTINE FIXP(LUCAT)
00102 2* COMMON /F(3,90),TR(3,90),PI(3,90),PF(3,90),V(3,90),F(3,90),
00103 3* 0(3,90),S(3,90),b(3,90),WT(3,90),FLM(3,90),FLV(3,90),FLMD,
00104 4* DELP(90),DELT(90),DELH(90),ETATH(90),ETAS(90),
00105 5* PRES(50),TEMP(50),VOL(50),FGAS(50),WIDTH(50),DIST(50),
00106 6* EFF(50),DEL(4),CP(4),CV,RWORK,RHEAT,Y,TREF,PCOEF,
00107 7* SLIP(50),BETA(50),RG(50),THICK,VLOAD,COND,VEL,VOLT,WRAT,SL
00108 8* DP=PIF(1,LOCAT+1)/PI(1,LOCAT)**(1/39.)
00109 9* EFF(1)=VLOAD
00110 10* SLIP(1)=SL
00111 11* L=LOCAT
00112 12* BETA(1)=F(1,L)/(F(1,L)+SL*(1-F(1,L)))
00113 13* RG(1)=L
00114 14* PRES(1)=PI(1,LOCAT)
00115 15* TEMP(1)=TR(1,LOCAT)
00116 16* FGAS(1)=F(1,LOCAT)
00117 17* VOL(1)=V(1,LOCAT)
00118 18* WIDTH(1)=W(1)
00119 19* DIST(1)=0.
00120 20* DP=22*(2.90+1)
00121 21* TREF(1)=TREF(1)-TEMP(1)-TEMP(1)*DP**Y)*EFF(1-1)
00122 22* PRES(1)=PRES(1)-DP
00123 23* VOL(1)=RWORK*TEMP(1)/PRES(1)
00124 24* VRAT=VOL(1)/V(1,LOCAT)
00125 25* WRAT=WT(1,LOCAT)/WT(3,LOCAT)
00126 26* FGAS(1)=WRAT*VRAT/(1+(VRAT-1)*WRAT)
00127 27* CALL EF1ZNC(1,L)
00128 28* EFF(1)=VLOAD*EFF(1)
00129 29* CALL EQUATE(L,LOCAT)
00130 30* RETURN
00131 31* END
  
```

END OF COMPILATION: NO DIAGNOSTICS.

SUBROUTINE FIX1 ENTRY POINT 00017

STORAGE USED: CODE(1) 000135; DATA(0) 000024; BLANK COMMON(2) 000164

EXTERNAL REFERENCES (BLOCK, NAME)

0003 EF1ZNC  
 0004 EQUATE

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000041	1216	0002	004160	B	0002	R	010011	DLTA	0002	010157	CUND	0002	007715	CP				
0002	007721	CV	0002	006535	DELM	0002	R	006251	DELTA	0002	006403	DELT	0002	R	007545	DISI			
0000	000000	DT	0000	000003	DTNS	0002	R	007627	EFF	0002	007021	ETAS	0002	006467	ETATH				
0002	002506	F	0002	R	007401	FGAS	0002	R	005214	FLM	0002	006250	FLHU	0002	005632	FLV			
0002	003124	H	0000	I	000001	I	0002	R	007726	PCOEF	0002	R	001452	PF	0002	001034	PI		
0002	R	007153	PRES	0002	R	010073	RG	0002	R	007723	RHEAT	0002	R	007722	RWORK	0002	003542	S	
0002	R	010163	SL	0002	R	007727	SLIP	0002	R	007235	TLHP	0002	R	000000	TF	0002	010155	THICK	
0002	R	008416	TR	0002	R	007725	TREF	0002	R	002070	V	0002	R	010160	VEL	0002	R	010156	VLOAD
0002	R	007317	VOL	0002	R	010161	VOLT	0000	R	000002	VKAT	0002	R	007711	WIDE	0002	R	007463	WIDTH
0002	R	010162	WRAT	0002	R	004576	WT	0002	R	007724	Y								

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00101 1* SUBROUTINE FIX1(L) 000000
00102 2* COMMON /F(1,90),TR(3,90),PI(3,90),PF(3,90),V(3,90),F(3,90), 000000
00103 3* (13,90),S(3,90),b(3,90),WT(3,90),FLM(3,90),FLV(3,90),FLMC, 000000
00103 4* DLEP(90),DELT(90),DELR(90),ETATH(90),ETAS(90), 000000
00103 5* PRES(50),TEMP(50),VOL(50),FGAS(50),WIDTH(50),D15T(50), 000000
00103 6* EFF(50),WIDE(4),CP(4),CV,RWORK,RHEAT,T,TREF,PCOEF, 000000
00103 7* SLIP(50),BETA(50),RG(50),THICK,VLOAD,CUND,VEL,VOLT,WRAT,SL 000000
00104 8* DT=(TR(1,L)+1)-TR(1,L))/39. 000000
00105 9* PRES(1)=PF(1,L) 000005
00106 10* TEMP(1)=TR(1,L) 000007
00107 11* VOL(1)=V(1,L) 000011
00110 12* WIDTH(1)=WIDE(1) 000013
00111 13* D15T(1)=0 000015
00112 14* FGAS(1)=F(1,L) 000016
00113 15* EFF(1)=VLOAD 000020
00114 16* SLIP(1)=SL 000022
00115 17* BETA(1)=F(1,L)/(F(1,L)+SL*(1-F(1,L))) 000024
00116 18* RG(1)=b 000032
00117 19* WKAT=WT(1,L)/WT(3,L) 000033
00120 20* DU=0.240,1 000041
00123 21* PRES(1)=PRES(1-1)*(1+DT/Y)/(TEMP(1-1)*EFF(1-1)) 000043
00124 22* TEMP(1)=TEMP(1-1)+DT 000053
00125 23* VOL(1)=RWORK*TEMP(1)/PRES(1) 000054
00126 24* VRAT=VOL(1)/V(1,L) 000061
00127 25* FGAS(1)=WRAT*VRAT/(1+(VKAT-1)*WRAT) 000063
00130 26* CALL EF1ZNC(1,L) 000072
00131 27* EFF(1)=VLOAD*EFF(1) 000075
00133 28* CALL EQUATE(L) 000102
00134 29* RETURN 000104
00135 30* END 000134
  
```

END OF COMPILATION# NO DIAGNOSTICS.

SUBROUTINE FIXV ENTRY POINT 000045

STORAGE USED: CODE(1) 000055; DATA(0) 000010. BLANK COMMON(2) 010164

EXTERNAL REFERENCES (BLOCK, NAME)

0003 FIXV

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000017	032L	0002	004160	6	0002	010011	BETA	0002	010157	COND	0002	007715	CP
0002	007721	CV	0002	006535	DELH	0002	006251	DELP	0002	006403	DELT	0002	007545	DIST
0000	000003	0YNS	0002	007627	EFF	0002	007021	ETAS	0002	006667	ETATH	0002	002506	F
0000	000001	FF	0002	007401	FGAS	0002	005214	FLM	0002	006250	FLMU	0002	005432	FLV
0002	003124	H	0002	007726	PCOEF	0002	001452	PF	0002	001034	PI	0002	007153	PRES
0002	010073	HG	0002	007723	RHEAT	0002	007722	RWORK	0002	003542	S	0002	010163	SL
0002	007727	SLIP	0000	000002	TEMP	0002	007235	TEMP	0002	000000	TF	0002	010155	THICK
0002	000416	TR	0002	007725	TREF	0002	002070	V	0002	010160	VEL	0002	010156	VLOAD
0002	007317	VOL	0002	010161	VOLT	0000	000000	VV	0002	007711	WIDE	0002	007463	WIDTH
0002	010162	WRAT	0002	004576	WT	0002	007724	Y						

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00101 1* SUBROUTINE FIXV(LOCAT) 000000
00102 2* COMMON /F(13,90),TR(13,90),PI(13,90),PF(13,90),Y(13,90),F(13,90), 000000
00103 3* I H(13,90),S(13,90),U(13,90),WT(13,90),FLM(13,90),FLV(13,90),FLMU, 000000
00103 4* DELP(90),DELT(90),DELH(90),ETATH(90),ETAS(90), 000000
00103 5* PRES(50),TEMP(50),VOL(50),FGAS(50),RIDTH(50),DIST(50), 000000
00103 6* EFF(50),WIDE(4),CP(4),CV,RWORK,RHEAT,Y,TREF,PCOEF, 000000
00103 7* SLIP(50),BETA(50),R6(50),THICK,VLOAD,COND,VEL,VOLT,WRAT,SL 000000
00104 8* VV =V(2,LOCAT)+I*(1-WT(1,LOCAT))*F(1,LOCAT+1)/((1-F(1,LOCAT+1)) 000000
00104 9* *WT(1,LOCAT)) 000000
00105 10* FF=F(1,LOCAT+1) 000012
00105 11* TEMP(40)=TR(1,LOCAT) 000014
00107 12* H32 TEM*TEMP(40) 000017
00110 13* PF(1,LOCAT+1)=RWORK*TEM/VV 000020
00111 14* CALL FIXP(LOCAT) 000025
00112 15* IF (ABS(1-FF-F(1,LOCAT+1))/FF)-.001834,034,632 000027
00115 16* H34 CONTINUE 000040
00116 17* RETURN 000040
00117 18* END 000054
  
```

END OF COMPILATION: NO DIAGNOSTICS.

WFOK.15 RHO.KEAT  
 FOR 0126-04.17775-13:1759 1.01

SUBROUTINE HLAT ENTRY POINT 000056

STORAGE USED: CODE(1) 000073; DATA(0) 000012; BLANK COMMON(2) 010164

INTERNAL REFERENCES (BLOCK, LABEL)

0003 VOLUN

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000034	1076	0004	004160	B	0002	010011	BETA	0002	010157	COND	0002	R	007715	CP	
0002	007721	CV	0004	006535	DELH	0002	006251	DELP	0002	006403	DELT	0002		007545	DIST	
0000	R	000600	0004	008002	BYR3	0002	007627	EFF	0002	007021	ETAS	0002		006667	ETATH	
0002	002506	F	0002	007401	FGAS	0002	005214	FLH	0002	006250	FLMO	0002		005632	FLV	
0002	003124	H	0000	1	000001	J	0002	007726	FLOLF	0002	R	001452	PF	0002	001034	P1
0002	007153	PKES	0002	010073	RG	0002	007723	RHEAT	0002	007722	RWORK	0002		003542	S	
0002	010163	SL	0002	007727	SLIP	0002	007235	TEMP	0002	000000	TF	0002		010155	THICK	
0002	R	000416	0002	007725	TREF	0002	R	002070	V	0002	010160	VEL	0002	010156	VLOAD	
0002	007317	VOL	0002	010161	VOLT	0002	007711	WIDE	0002	007463	WIDTH	0002		010162	WRAT	
0002	R	004576	0002	007724	Y											

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00101 1* SUBROUTINE HEAT(L1,L2,OH,WP)
00102 2* COMMON /ET(3,90),TR(3,90),H(3,90),PF(3,90),V(3,90),F(3,90),
00103 3* H(3,90),S(3,90),p(3,90),WT(3,90),FLN(3,90),FLV(3,90),FLMO,
00104 4* DELP(90),DELT(90),DELH(90),ETATH(90),ETAS(90),
00105 5* FES(50),TEMP(50),VCL(50),FGAS(50),WIDTR(50),DIST(50),
00106 6* EFF(50),WIDL(4),CP(4),CV,RWORK,RHEAT,Y,TREF,PCOEF,
00107 7* SLIP(50),BETA(50),RG(50),THICK,VLOAD,LOND,VEL,VOLT,WRAT,SL
00108 8*
00109 9* DP=WP*PF(3,L1)*0.01284
00110 10* U1=(OH-V(2,L1)*WT(2,L1)*DP)/(WT(1,L1)*CP(1)+WT(2,L1)*CP(2))
00111 11* DO 000 J=1,3,1
00112 12* TR(J,L2)=TR(J,L1)+U1
00113 13* HBU AT(J,L2)=AT(J,L1)
00114 14* CALL VOLUN(L2)
00115 15* RETURN
00116 16* END
00004
00004
00004
00004
00004
00004
00004
00004
00004
00010
00034
00034
00036
00041
00044
00046
00072

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END OF COMPILATION: NO DIAGNOSTICS.

SUBROUTINE REPORT ENTRY POINT 000067

STORAGE USED: CODE(1) 000072; DATA(0) 000257; BLANK COMMON(2) 007726

LATERAL REFERENCES (BLOCK, NAME)

0003 NIDCBS  
 0004 NSCKs  
 0005 RNFUS  
 0006 RNFUS

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0000	000135	1200	0000	000141	1220	0000	000150	1270	0000	000157	1340	0000	000164	1410
0000	000175	1460	0000	000204	1530	0000	000213	1600	0000	000222	1650	0000	000231	1720
0000	000240	1770	0000	000247	2040	0000	000000	502F	0000	000002	509F	0000	000011	506F
0000	000021	508F	0000	000027	510F	0000	000041	512F	0000	000054	514F	0000	000064	516F
0000	000075	518F	0000	000104	520F	0000	000114	522F	0000	000125	524F	0002	R	004160
0002	007715	CP	0002	007721	CV	0002	006535	DELH	0002	006251	DELP	0002	R	006403
0002	007545	DIST	0000	000253	DYNS	0002	007627	EFF	0002	007021	ETAS	0002	R	006667
0002	R	002506	F	0002	007401	FGAS	0002	R	005214	FLM	0002	R	006450	FLM
0002	R	003124	H	0002	001452	PF	0002	R	001034	PI	0002	R	007153	PRES
0002	R	007722	RWORK	0002	R	003542	S	0002	R	007235	TEMP	0002	R	000000
0002	R	007725	TRLF	0002	R	002070	V	0002	R	007317	VOL	0002	R	007711
0002	R	004576	BT	0002	R	007724	Y	0002	R	007711	WIDE	0002	R	007463

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00101 1+ SUBROUTINE REPORT(L)
00103 2+ COMMON TF(3,90),TR(3,90),PI(3,90),PF(3,90),V(3,90),F(3,90),
00103 3+ H(3,90),S(3,90),B(3,90),WT(3,90),FLM(3,90),FLV(3,90),FLHD,
00103 4+ DLLP(90),DELI(90),DELH(90),ETATH(90),ETAS(90),
00103 5+ PRES(50),TEMP(50),VOL(50),FGAS(50),WIDTH(50),DIST(50),
00103 6+ EFF(50),WIDE(14),CP(4),CV,RWORK,RHEAT,Y,TRF
00104 7+
502 FORMAT('0')
00105 8+
504 FORMAT(15,'GAS',16,'LIQUID',17,'TOTAL')
00106 9+
506 FORMAT('0',12,'TEMPERATURE,DEG.F',T51,3F12.2)
00107 10+
508 FORMAT(12,'PRESSURE,PSI',T51,3F12.2)
00110 11+
510 FORMAT(12,'MASS FRACTION,REFERRED TO MHD DUCT INLET',T51,3F12.6)
00111 12+
512 FORMAT(12,'ENTHALPY,BTU PER POUND TOTAL AT MHD DUCT INLET',
00111 13+ T51,3F12.3)
00112 14+
514 FORMAT(12,'ENTROPY,BTU/R-LB(INLET MASS)',T51,3F12.4)
00113 15+
516 FORMAT(12,'AVAILABILITY,BTU/LB(INLET MASS)',T51,3F12.4)
00114 16+
518 FORMAT(12,'VOLUME FRACTION,LOCAL',T51,3F12.4)
00115 17+
520 FORMAT(12,'SPECIFIC VOLUME,CUBIC FEET PER POUND',T51,3F12.4)
00116 18+
522 FORMAT(12,'MASS FLOW RATE, LB/SEC',T51,3F12.2)
00117 19+
524 FORMAT(12,'VOLUME FLOW RATE,CU-FT/SEC',T51,3F12.1)
00120 20+ WRITE(6,504)
00122 21+ WRITE(6,506)TF(1,L),TF(2,L),TF(3,L)
00127 22+ WRITE(6,508)PI(1,L),PI(2,L),PI(3,L)
00134 23+ WRITE(6,510)WT(1,L),WT(2,L),WT(3,L)
00141 24+ WRITE(6,512)H(1,L),H(2,L),H(3,L)
00146 25+ WRITE(6,514)S(1,L),S(2,L),S(3,L)
00153 26+ WRITE(6,516)B(1,L),B(2,L),B(3,L)
00160 27+ WRITE(6,518)F(1,L),F(2,L),F(3,L)
00165 28+ WRITE(6,520)V(1,L),V(2,L),V(3,L)
00172 29+ WRITE(6,522)FLM(1,L),FLM(2,L),FLM(3,L)
00177 30+ WRITE(6,524)FLV(1,L),FLV(2,L),FLV(3,L)
00204 31+ WRITE(6,502)
00206 32+ RETURN
00207 33+ END
  
```

END OF COMPILATION: NO DIAGNOSTICS.

000071



SUBROUTINE SEPER ENTRY POINT 000106

STORAGE USED: CODE(1) 000120; DATA(0) 000041; BLANK COMMON(2) 010164

EXTERNAL REFERENCES (BLOCK, NAME)

0003 VOLUM

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000007	1166	0001	000037	1146	0001	000050	1226	0002	004160	B	0002	010011	BETA
0002	010157	COND	0002	007715	CP	0002	007721	CV	0002	006535	DELH	0002	006251	DELP
0002	006403	DELT	0002	007545	DIST	0000	000020	DYNS	0002	007427	EEF	0002	007021	ETAS
0002	006667	ETATH	0002	002506	F	0000	000003	FC	0002	007401	FGAS	0002	005214	FLN
0002	006250	FLHD	0002	005632	FLV	0000	000000	FS	0002	003124	H	0000	000016	J
0000	000014	L	0000	000015	LL	0000	000017	LLL	0002	007726	PCOEF	0002	001452	PF
0002	001034	PI	0002	007153	PRES	0002	010073	RG	0002	007723	RHEAT	0002	007722	RWRK
0002	003542	S	0002	010163	SL	0002	007727	SLIP	0002	007235	TEMP	0002	000000	TE
0002	010155	THICK	0002	R 000416	TR	0002	007725	TREF	0002	002070	V	0002	010160	VEL
0002	010156	VLOAD	0002	007317	VOL	0002	010161	VOLT	0002	007711	WIDE	0002	007463	WIDTH
0002	010162	WRAT	0002	R 004576	WT	0002	007724	Y						

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00101 1* SUBROUTINE SEPER (LOCAT, FS1, FS2) 000000
00103 2* COMMON /FT(3,90), TR(3,90), P(3,90), PF(3,90), V(3,90), F(3,90), 000000
00103 3* H(3,90), S(3,90), B(3,90), WT(3,90), FLH(3,90), FLV(3,90), FLHD, 000000
00103 4* DELP(90), DELT(90), DELH(90), ETATH(90), ETAS(90), 000000
00103 5* PRES(50), TEMP(50), VOL(50), FGAS(50), WIDTH(50), DIST(50), 000000
00103 6* EEF(50), WIDE(4), CPI(4), CV, RWRK, RHEAT, Y, TREF, PCOEF, 000000
00103 7* SLIP(50), BETA(50), RG(50), THICK, VLOAD, COND, VEL, VOLT, WRAT, SL 000000
00104 8* DIMENSION FS(3), FC(3,3) 000000
00105 9* FS(1)=FS1 000000
00106 10* FS(2)=FS2 000001
00107 11* DO 406 L=1,2,1 000007
00112 12* LL=LOCAT+L-1*10 000020
00113 13* DO 402 J=1,3,1 000025
00116 14* TR(J,LL+1)=TR(J,LL) 000037
00117 15* PF(J,LL+1)=PF(J,LL) 000040
00121 16* DO 404 J=1,2,1 000050
00124 17* FC(J,L)=FS(J)*(3-2*L)+L-1 000050
00125 18* 404 WT(J,LL+1)=WT(J,LL)+FC(J,L) 000054
00127 19* *1(3,LL+1)=*1(1,LL+1)+*1(2,LL+1) 000060
00130 20* LLL=LL+1 000064
00131 21* 406 CALL VOLUR(LLL) 000067
00133 22* RETURN 000074
00134 23* END 000117
  
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END OF COMPILATION: NO DIAGNOSTICS.

SUBROUTINE SHUNT ENTRY POINT 000077

STORAGE USED: CODE(1) 00014; DATA(0) 000024; BLANK COMMON(2) 010164

EXTERNAL REFERENCES (BLOCK, NAME)

0003 VOLUN  
 0004 EXP  
 0005 ALOG

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000052	116	0002	004160	B	0002	010011	BETA	0002	R	010157	COND	0002	R	007715	CR		
0002	007721	CV	0002	006535	DELH	0002	006251	DELP	0002	R	006403	DELT	0002		007545	DIST		
0000	000003	DYNS	0002	007627	EFF	0002	007021	ETAS	0002		006667	ETATH	0002	R	002506	F		
0002	007401	FGAS	0002	005214	FLM	0002	006250	FLMU	0002		005632	FLY	0002		003124	H		
0000	000002	J	0002	007726	PCOEF	0002	R	001452	PF	0002		001034	PI	0000	R	000000	PIE	
0002	007153	PRES	0000	R	000001	RESIS	0002	010073	RG	0002		007723	RHEAT	0002		007722	RWORK	
0002	003542	S	0002	010123	SL	0002	007727	SLIP	0002		007235	TEMP	0002		000000	TR		
0002	010155	THICK	0002	R	000416	TR	0002	007725	TREF	0002	R	002070	Y	0002	R	010160	VEL	
0002	R	010156	VLOAD	0002	007317	VOL	0002	R	010161	VOLT	0002		007711	WIDE	0002		007463	WIDTH
0002	010162	WRAT	0002	R	004576	W1	0002	007724	Y									

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00101 1* SUBROUTINE SHUNT(L, WTH, FCT) 000007
00103 2* COMMON IF(3,90),TR(3,90),PI(3,90),FF(3,90),Y(3,90),F(3,90), 000007
00103 3* H(3,90),S(3,90),B(3,90),WT(3,90),FLN(3,90),ELY(3,90),Z(3,90),FLM, 000007
00103 4* DELP(90),DELT(90),DELH(90),ETATH(90),ETAS(90), 000007
00103 5* PRES(50),TEMP(50),VOL(50),FGAS(50),WIDTH(50),DIST(50), 000007
00103 6* EFF(50),WIDE(4),CP(4),CV,RWORK,RHEAT,Y,TREF,PCOEF, 000007
00103 7* SLIP(50),BETA(50),RG(50),THICK,VLOAD,COND,VEL,VOLT,WRAT,SL 000007
00104 8* PIE=3.14159 000007
00105 9* RESIS=EXP(3.8*F(1,1))/COND 000011
00106 10* DELH(L)=ALOG(2)*VOLT*V(3,L)*FCT*.000946*VLOAD*VLOAD/ 000020
00106 11* (PIE*RESIS*WTH*VEL) 000020
00107 12* DELT(L)=DELH(L)/(WT(1,L)*LP(1)+WT(2,L)*CP(2)) 000037
00110 13* DO 002 J=1,3,1 000052
00113 14* TR(J,L+1)=TR(J,L)+DELT(L) 000052
00114 15* PI(J,L+1)=PI(J,L) 000054
00115 16* 602 WT(J,L+1)=WT(J,L) 000056
00117 17* CALL VOLUN(L+1) 000061
00120 18* RETURN 000066
00121 19* END 000113
  
```

END OF COMPILATION: NO DIAGNOSTICS.

SUBROUTINE SINK ENTRY POINT 000065

STORAGE USED: CODE(1) 000102; DATA(1) 000017; BLANK COMMON(2) 010164

EXTERNAL REFERENCES (BLOCK, NAME)

0003 VOLUM  
 0004 ALOG

STORAGE ASSIGNMENT (BLOCK, TYPL, RELATIVE LOCATION, NAME)

0001	000023	1650	0002	004160	B	0002	010011	BETA	0002	010157	COND	0002	007715	CP
0002	007721	CV	0002	006535	DELH	0002	006251	DELPL	0002	006403	DELTT	0002	007545	DIST
0000	000002	DYNS	0002	007827	EFF	0002	007021	ETAS	0002	R 006667	ETATH	0002	002506	F
0002	007401	FGAS	0002	005214	FLH	0002	006250	FLMO	0002	005632	FLV	0002	003124	H
0000	000000	J	0002	007726	PCOEF	0002	R 001452	PF	0002	001034	PI	0002	007153	PRES
0002	010073	RG	0002	007723	RHEAT	0002	007722	RWORK	0002	003592	S	0002	010163	SL
0002	007727	SLIP	0002	007235	TEMP	0002	000000	TF	0002	010155	THICK	0002	R 000416	TR
0002	n 007725	TREF	0002	n 000001	TAX	0002	002070	Y	0002	010160	VEL	0002	010156	VLOAD
0002	007317	VOL	0002	010161	VOLT	0002	007711	WIDE	0002	007463	WIDTH	0002	010162	WRAT
0002	R 004576	WT	0002	007724	Y									

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00101	10	SUBROUTINE SINK(L1,L2,L2,DP)	000007
00102	20	COMMON TR(3,90),PI(3,90),PF(3,90),V(3,90),F(3,90),	000007
00103	30	H(3,90),S(3,90),B(3,90),WT(3,90),FLH(3,90),FLV(3,90),FLMO,	000007
00104	40	DELPL(90),DELTT(90),DELH(90),ETATH(90),ETAS(90),	000007
00105	50	PRES(50),TEMP(50),VOL(50),FGAS(50),WIDTH(50),DIST(50),	000007
00106	60	EFF(50),WIDE(4),CP(4),CV,RWORK,RHEAT,Y,TREF,PCOEF,	000007
00107	70	SLIP(50),RHEA(50),RG(50),THICK,VLOAD,COND,VEL,VOLT,WRAT,SL	000007
00108	80	DO 902 J=1,3,1	000007
00109	90	TR(J,L2)=12	000023
00110	100	PF(J,L2)=PF(J,L1)*(1+DP)	000024
00111	110	WT(J,L2)=WT(J,L1)	000027
00112	120	CALL VOLUM(L2)	000032
00113	130	TAX=TR(1,L1)*0.001	000036
00114	140	ETATH(L1)=1-TREF*ALOG(TAX /TR(1,L2))/TAX	000041
00115	150	RETURN	000054
00116	160	END	000103

END OF COMPILATION: NO DIAGNOSTICS.

MFUR, IS HHD, SKIP  
 FOR DT20-04/17775-13:10:21 (,0)

SUBROUTINE SKIP ENTRY POINT 000046

STORAGE USED: CODE(1) 000047; DATA(0) 000014; BLANK COMMON(2) 010164

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000015	1056	0002	004160	B	0002	010011	BETA	0002	010157	COND	0002	007715	CP
0002	007721	CV	0002	006535	DELH	0002	006251	DELP	0002	004403	DELT	0002	007545	DIST
0000	000001	DYNS	0002	007627	EFF	0002	007021	ETAS	0002	006667	ETATH	0002 R	002504	F
0002	007401	FGAS	0002	005214	FLH	0002	006250	FLHU	0002	005632	FLV	0002	003124	H
0000	000000	J	0002	007726	KCOEF	0002 K	001452	PF	0002	001034	PI	0002	007153	PRES
0002	010073	K6	0002	007723	KHEAT	0002	007722	RWORK	0002	003542	S	0002	010163	SL
0002	007727	SLIP	0002	007235	TEMP	0002	000000	TF	0002	010155	THICK	0002 K	000416	TR
0002	007725	TREF	0002 K	002070	V	0002	010160	VEL	0002	010156	VLOAD	0002	007317	VOL
0002	010161	VOLT	0002	007711	WIDE	0002	007463	WIDTH	0002	010162	WRAT	0002 R	004576	WT
0002	007724	Y												

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00101 1* SUBROUTINE SKIP(L1,L2) 000015
00102 2* COMMON /3(3,90),TR(3,90),P1(3,90),PF(3,90),V(3,90),F(3,90), 000015
00103 3* H(3,90),S(3,90),B(3,90),WT(3,90),FLH(3,90),FLV(3,90),FLHO, 000015
00104 4* DELP(90),DELT(90),DELH(90),ETATH(90),ETAS(90), 000015
00105 5* PRES(50),TEMP(50),VOL(50),FGAS(50),WIDTH(50),DIST(50), 000015
00106 6* EFF(50),WIDE(4),CP(4),CV,RWORK,RHEAT,Y,TREF,KCOEF, 000015
00107 7* SLIP(50),BETA(50),RG(50),THICK,VLOAD,COND,VEL,VOLT,WRAT,SL 000015
00108 8* DO 302 J=1,3,1 000015
00109 9* TR(J,L2)=TR(J,L1) 000015
00110 10* PF(J,L2)=PF(J,L1) 000016
00111 11* V(J,L2)=V(J,L1) 000020
00112 12* WT(J,L2)=WT(J,L1) 000022
00113 13* F(J,L2)=F(J,L1) 000024
00114 14* RETURN 000027
00115 15* END 000046

```

END OF COMPILATION: NO DIAGNOSTICS.

SUBROUTINE THROT ENTRY POINT 000854

STORAGE USED: CODE(1) 000073; DATA(1) 000013; BLANK COMMON(2) 010164

EXTERNAL REFERENCES (BLOCK, NAME)

0003 VOLUN

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	00020	1000	0002	009160	B	0002	010011	BETA	0002	010157	COND	0002	R	007715	CE	
0002	007721	CV	0002	006555	DELH	0002	006251	DELP	0002	006403	DELT	0002		007545	DIST	
0000	000001	DYN	0002	007627	EFF	0002	007071	ETAS	0002	006667	ETATH	0002		002506	F	
0002	007401	FAS	0002	005414	FLH	0002	006250	FLMO	0002	005632	FLV	0002		003124	H	
0000	000000	T	0002	007726	PCOEF	0002	R	001452	PF	0002	001034	PI	0002		007153	PRES
0002	010073	RG	0002	007723	RHEAT	0002	007722	RWORK	0002	003542	S	0002		010163	SL	
0002	007727	SLIP	0002	007235	TEMP	0002	000000	TF	0002	010155	THICK	0002	R	000416	TH	
0002	007725	TREF	0002	R	002070	V	0002	010160	VEL	0002	010156	VLOAD	0002		007317	VOL
0002	010161	VOLT	0002	007711	WIDE	0002	007463	WIDTH	0002	010162	WRAT	0002	R	004576	WT	
0002	007724	Y														

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00101      1*      SUBROUTINE THROT(L1,L2,DP)      000004
00103      2*      COMMON /FT(3,90),TR(3,90),P(3,90),PF(3,90),V(3,90),F(3,90),  000004
00105      3*      1      H(3,90),S(3,90),B(3,90),WT(3,90),FLH(3,90),FLV(3,90),FLMO,  000004
00107      4*      2      DELP(90),DELT(90),DELH(90),ETATH(90),ETAS(90),  000004
00109      5*      3      PLES(50),TEMP(50),VOL(50),FAS(50),WIDTH(50),DIST(50),  000004
00111      6*      4      EFF(50),WIDE(4),CP(4),CV,RWORK,RHEAT,Y,TREF,PCOEF,  000004
00113      7*      5      SLIP(50),BETA(50),RG(50),THICK,VLOAD,COND,VEL,VOLT,WRAT,SL  000004
00115      8*      DP=DP+PF(1,L1)      000004
00117      9*      DO 842 J=1,3,1      000007
00119     10*      WT(J,L2)=WT(J,L1)      000020
00121     11*      PF(J,L2)=PF(J,L1)+DP      000021
00123     12*      842 TR(J,L2)=TR(J,L1)+WT(2,L1)*V(2,L1)*DP+.001264/  000024
00125     13*      1      (WT(2,L1)*CP(2)+WT(1,L1)*CP(1))  000024
00127     14*      CALL VOLUN(L2)      000041
00129     15*      RETURN      000043
00131     16*      END      000072
  
```

END OF COMPILATION: NO DIAGNOSTICS.

SUBROUTINE VALUE ENTRY POINT 000152

STORAGE USED: CODE(1) 000165; DATA(0) 000024; BLANK COMMON(2) 010164

EXTERNAL REFERENCES (BLOCK, NAME)

0003 VOLUM  
 0004 ALOG

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000013	1066	0001	000104	1256	0001	000133	1366	0002	R	004160	B	0004	010011	BETA		
0002	010157	CORR	0002	K	007715	CP	0002	007721	CV	0002	006535	DELH	0002	006251	DELP		
0002	006403	DELT	0002	007545	DIST	0000	000003	DYNS	0002	007627	EFF	0002	007021	ETAS			
0002	006607	ETATH	0002	002506	F	0002	007401	FGAS	0002	005214	FLM	0002	006250	FLMO			
0002	005632	FLV	0002	K	003124	H	0000	1	000001	J	0002	007726	PCOEF	0002	K	001452	PF
0002	K	001034	PI	0000	K	000000	PREF	0002	007153	PRES	0002	010073	RG	0002	K	007723	RHEAT
0002	007722	RWORK	0002	K	003542	S	0002	010163	SL	0002	007727	SLIP	0000	K	000002	TAB	
0002	007235	TEHP	0002	K	000000	TF	0002	010155	THICK	0002	R	000416	TR	0002	R	007725	TREF
0002	K	002070	V	0002	010160	VEL	0002	010156	VLOAD	0002	007317	VOL	0002	010161	VOLT		
0002	007711	WIDE	0002	007443	WIDTH	0002	010162	WRAT	0002	R	004576	WT	0002	007724	Y		

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00101      1*      SUBROUTINE VALUE(L)
00102      2*      COMMON I(3,90),IK(3,90),PI(3,90),PF(3,90),V(3,90),F(3,90),
00103      3*      H(3,90),S(3,90),B(3,90),WT(3,90),FLM(3,90),FLV(3,90),FLMO,
00104      4*      DELP(90),DELT(90),DELH(90),ETATH(90),ETAS(90),
00105      5*      PRES(50),TEHP(50),VOL(50),FGAS(50),WIDTH(50),DIST(50),
00106      6*      EFF(50),WIDE(4),CP(4),CV,RWORK,RHEAT,Y,TREF,PCOEF,
00107      7*      SLIP(50),BETA(50),RG(50),THICK,VLOAD,COND,VEL,VOLT,WRAT,SL
00108      8*      PREF=14.7*144.
00109      9*      DO 742 J=2,3,1
00110     10*      TR(J,L)=TR(1,L)
00111     11*      742 PF(J,L)=PF(1,L)
00112     12*      CALL VOLUM(L)
00113     13*      S(3,L)=0
00114     14*      B(3,L)=0
00115     15*      H(3,L)=0
00116     16*      H(2,L)=WT(2,L)*V(2,L)*(PF(2,L) )*.001284
00117     17*      H(1,L)=0
00118     18*      TAB=959.69
00119     19*      S(1,L)=W(1,L)*(CP(1)*ALOG(TR(1,L)/TAB) -RHEAT*ALOG(PF(1,L)/PREF))
00120     20*      S(2,L)=WT(2,L)*WT(2,L)*CP(2)*ALOG(TR(2,L)/TAB)
00121     21*      DO 744 J=1,2,1
00122     22*      H(J,L)=WT(J,L)*CP(J)*(TR(J,L)-TREF)*H(J,L)
00123     23*      B(J,L)=H(J,L)-TREF*S(J,L)
00124     24*      S(3,L)=S(3,L)+S(J,L)
00125     25*      H(3,L)=H(3,L)+H(J,L)
00126     26*      744 H(3,L)=S(3,L)+B(J,L)
00127     27*      DO 746 J=1,3,1
00128     28*      PJ(J,L)=PF(J,L)*144
00129     29*      746 TF(J,L)=TR(J,L)-459.69
00130     30*      RETURN
00131     31*      END
000000
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000037
000061
000104
000104
000112
000116
000121
000123
000133
000133
000141
000141
000164

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END OF COMPILATION: NO DIAGNOSTICS.

WFOR.15      HND.VOLUM  
 FOR DT28-04/1775-13:18:36 (LUI)

SUBROUTINE VOLUM      ENTRY POINT 000062

STORAGE USED: CODE(1) 000070; DATA(0) 000014; BLANK COMMON(2) 010164

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000045	1146	0002	004160	B	0002	010011	BLTA	0002	010157	COND	0002	007715	CR			
0002	007721	CV	0002	006535	DELH	0002	006251	DELTA	0002	006403	DELT	0002	007545	DIST			
0000	000002	DYNS	0002	007627	EFF	0002	007021	ETAS	0002	006667	ETATH	0002	R	002506	F		
0002	007901	FGAS	0002	R	0065214	FLM	0002	R	006250	FLMU	0002	R	006632	FLV	0002	003124	H
0000	000001	J	0002	007726	PCOEF	0002	R	001452	PF	0002	001034	PI	0002	007153	PRES		
0002	010073	RG	0002	007723	RHEAT	0002	R	007722	RWORK	0002	003542	S	0002	010163	SL		
0002	007727	SLIP	0002	007235	TEMP	0002	R	000000	T	0002	010155	THICK	0002	R	000416	TR	
0002	007725	TREF	0002	R	002078	V	0002	010160	VEL	0002	010156	VLOAD	0002	007317	VOL		
0002	010161	VOLI	0000	R	000000	VRAT	0002	007711	WIDE	0002	007463	WIDTH	0002	R	010162	WHAT	
0002	R	004576	WT	0002	007724	Y											

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00101      1*      SUBROUTINE VOLUM(LOCAT)      000006
00103      2*      COMMON IF(3,90),TK(3,90),PI(3,90),PF(3,90),V(3,90),F(3,90),      000006
00103      3*      1      HI(3,90),S(3,90),B(3,90),WT(3,90),FLH(3,90),FLV(3,90),FLMO,      000006
00103      4*      2      DELP(90),DELT(90),DELH(90),ETATH(90),ETAS(90),      000006
00103      5*      3      PRES(50),TEMP(50),VOL(50),FGAS(50),WIDTH(50),DIST(50),      000006
00103      6*      4      EFF(50),WIDE(4),CPI(4),CV,RWORK,RHEAT,Y,TREF,PCOEF,      000006
00103      7*      5      SLIP(50),BETA(50),RG(50),THICK,VLOAD,COND,VEL,VOLI,WHAT,SL      000006
00104      8*      V(1,LOCAT)=RWORK*TK(1,LOCAT)/PF(1,LOCAT)      000006
00105      9*      V(3,LOCAT)=V(1,LOCAT)*WT(1,LOCAT)+V(2,LOCAT)*WT(2,LOCAT)      000012
00106      10*      1      F(1(3,LOCAT))      000012
00107      11*      VRAT=V(1(1,LOCAT))/V(2,LOCAT)      000020
00107      12*      WRAT=V(1(1,LOCAT))/V(1(3,LOCAT))      000023
00110      13*      F(1(1,LOCAT))=VRAT*WRAT/(1+VRAT*WRAT)      000026
00111      14*      F(2,LOCAT)=1-F(1,LOCAT)      000035
00112      15*      F(3,LOCAT)=1      000037
00113      16*      DO 750 J=1,3,1      000045
00116      17*      FLH(J,LOCAT)=FLMO*WT(J,LOCAT)      000045
00117      18*      750      FLV(J,LOCAT)=FLH(J,LOCAT)*V(J,LOCAT)      000047
00121      19*      RETURN      000052
00122      20*      END      000067
  
```

END OF COMPILATION:      NO DIAGNOSTICS.

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Subappendix AA 11.2.2

SAMPLE OUTPUT



\*\*\*\*\*INPUT DATA

SPECIFIED MW INLET CONDITIONS

GAS MOL.WT= 19.95\*\*\*GAS CP= .1240 BTU/LB  
LIQ.DENS= 49.80 LB/CU FT\*\*\*LIQ.CP= .3000 BTU/LB\*\*\*LIQ.RESIS= .9660-05 OHM-CH

DUCT INLET SHUNT LOSS REDUCTION= .0000\*\*\*DUCT OUTLET SHUNT LOSS REDUCTION= .0000

COMPRESSOR POLYTROPIC EFFICIENCY= .8750\*\*\*LIQUID PUMP EFFICIENCY= .8500  
TEMP= 1200.00 DEG.F\*\*\*PRES= 1200.00 PSI\*\*\*VOID VOL= .6500

FRACT.OF TOTAL GAS SEPERATED IN GAS STREAM=1.0000\*\*\*FRACT.LIQUID IN GAS STREAM= .0000

SPEC.COMP.IN.TEMP= 1200.00 DEG.F\*\*\*COMP.EF= .8750\*\*\*CYCLE REJECT TEMP= 100.00 DEG.F

GAS STREAM PRESSURE LOSSES AS DP/P  
NOZZLE-SEP= .0200\*\*STEAM GEN.= .0200\*\*REGEN.LO PR SIDE= .0200\*\*FEED WATER HEATER .0000  
1ST HST. HT. X= .0200 2ND HST.HT. X= .0200 3RD HST.HT. X= .0200 4TH HST.HT. X= .0200  
REGEN.HI.SIDE= .0200\*\*PRIMARY HEATER= .0200\*\*MIXER= .0200

LIQUID STREAM PRESSURE LOSSES AS DP/P  
NOZZLE-SEP= .0500\*\*COOLER= .0000\*\*HEATER= .0400\*\*MIXER= .0000

ASSUMED ST.CYC.INT.EFFIC.FOR ST.GEN.= .7700 FOR FEED WATER HEATER .0000AND FOR PRECOOLERS .0000 .0000 .0000 .0000

STACK EFF.= .8800FRACT. OF PRECOOLER HEAT USED FOR AIR PREHEAT= 1.0000 .0000 .0000 .0000

FRACT. OF STEAM HEAT FOR DIRECT ST. REHEAT= .3500FROM 1000.0 TO 800.0 DEG.F.AT INTERNAL EFF. .8000

MIX.VEL= 100.00 FT/SEC\*\*INLET WIDTH= 5.85 FT\*\*THICKNESS= 5.85 FT

MAG.FLUX DEN= .00 TESLA\*\*LOADING= .9500 VOLT/VOLT\*\*INLET SLIP RATIO=1.00

OPEN CIR.VOLT= 33.00 WORKING VOLTAGE= .00

NO.COMP.STAGES= 1 \*NO.REGEN.= 0 \*REG.APP= .00 DEG.F \*REG.CUTOFF=150.00 DEG.F

COMB.CYC.POWER PER DUCT= 62.50MW, ITERATION INDEX= 1 COST INDEX= 1

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\*\*\*\*\*OUTPUT DATA-BASED ON ONE POUND MIXTURE AT MHD DUCT INLET

\*\*\*\*\*OUTLET FROM MIXER\*\*\*\*\* Point 1 of Fig. A 11.2.1

	GAS	LIQUID	TOTAL
TEMPERATURE, DEG. F	1200.00	1200.00	1200.00
PRESSURE, PSI	1200.00	1200.00	1200.00
MASS FRACTION, REFERRED TO MHD DUCT INLET			
ENTHALPY, BTU PER POUND TOTAL AT MHD DUCT INLET	.088771	.911229	1.000000
ENTROPY, BTU/N-LB(INLET MASS)	12.108	304.765	316.874
AVAILABILITY, BTU/LB(INLET MASS)	-.0053	.3198	.3145
VOLUME FRACTION, LOCAL	15.0780	125.7738	140.8518
SPECIFIC VOLUME, CUBIC FEET PER POUND	.650000	.350000	1.000000
MASS FLOW RATE, LB/SEC	.3828	.0201	.4029
VOLUME FLOW RATE, CU.FT/SEC	727.89	7471.68	8199.57
	278.6	150.0	428.6

\*\*\*\*\*INLET TO MHD DUCT\*\*\*\*\* Point 2 of Fig. A 11.2.1

	GAS	LIQUID	TOTAL
TEMPERATURE, DEG. F	1200.00	1200.00	1200.00
PRESSURE, PSI	1200.00	1200.00	1200.00
MASS FRACTION, REFERRED TO MHD DUCT INLET			
ENTHALPY, BTU PER POUND TOTAL AT MHD DUCT INLET	.088771	.911229	1.000000
ENTROPY, BTU/N-LB(INLET MASS)	12.108	304.765	316.874
AVAILABILITY, BTU/LB(INLET MASS)	-.0053	.3198	.3145
VOLUME FRACTION, LOCAL	15.0780	125.7738	140.8518
SPECIFIC VOLUME, CUBIC FEET PER POUND	.650000	.350000	1.000000
MASS FLOW RATE, LB/SEC	.3828	.0201	.4029
VOLUME FLOW RATE, CU.FT/SEC	727.89	7471.68	8199.57
	278.6	150.0	428.6

SPECIFIED MHD OUTLET VOID FRACTION= .8500

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IN DUCT	DIST.-FT	WIDTH-FT	TEMP-DEG.K	PRES-LB/SQFT	AREA FRAC.	GAS VOL-FRAC	GAS SLIP	RATIO	LOCAL EFFIC.
	.0000	2.0704	1659.6700	172800.0000	.6500	.6500	1.0000	.9500	
	.1987	1.8726	1658.9818	167871.5723	.6567	.6565	1.4608	.7410	
	.3974	1.6849	1658.4296	163083.7090	.6646	.6629	1.9554	.7410	
	.5961	1.5078	1657.9776	158432.4004	.6706	.6692	1.5223	.7410	
	.7948	1.3417	1657.6258	153913.7520	.6769	.6755	1.9271	.7410	
	.9935	1.1862	1656.7742	149523.9745	.6831	.6818	1.5318	.7410	
	1.1922	1.0419	1656.2227	145259.4062	.6893	.6880	1.9367	.7410	
	1.3909	.9082	1655.6715	141116.4629	.6954	.6941	1.5418	.7410	
	1.5896	.7850	1655.1204	137091.6816	.7014	.7001	1.9470	.7410	
	1.7883	.6718	1654.5695	133181.6914	.7075	.7061	1.5523	.7410	
	1.9870	.5686	1654.0188	129383.2178	.7134	.7120	1.9579	.7410	
	2.1857	.4754	1653.4683	125693.0801	.7193	.7178	1.5636	.7410	
	2.3844	.3922	1652.9180	122108.1885	.7252	.7236	1.9694	.7410	
	2.5831	.3190	1652.3678	118625.5420	.7309	.7292	1.5754	.7410	
	2.7818	.2558	1651.8178	115242.2236	.7367	.7349	1.9817	.7410	
	2.9805	.2026	1651.2681	111955.4019	.7423	.7409	1.5880	.7410	
	3.1792	.1594	1650.7184	108762.3223	.7479	.7458	1.9946	.7410	
	3.3779	.1262	1650.1690	105660.3135	.7535	.7512	1.6013	.7410	
	3.5766	.0930	1649.6198	102646.7764	.7590	.7565	1.6083	.7410	
	3.7753	.0698	1649.0707	99719.1885	.7644	.7618	1.6154	.7410	
	3.9740	.0466	1648.5219	96875.0986	.7697	.7669	1.6227	.7410	
	4.1727	.0334	1647.9732	94112.1250	.7750	.7720	1.6302	.7410	
	4.3714	.0202	1647.4247	91427.9591	.7802	.7770	1.6378	.7410	
	4.5701	.0170	1646.8763	88820.3379	.7854	.7819	1.6457	.7410	
	4.7688	.0138	1646.3282	86287.0937	.7905	.7867	1.6538	.7410	
	4.9675	.0106	1645.7802	83826.1066	.7955	.7915	1.6621	.7410	
	5.1662	.0074	1645.2325	81435.2969	.8005	.7962	1.6706	.7410	
	5.3649	.0042	1644.6849	79112.6816	.8053	.8008	1.6793	.7410	
	5.5636	.0010	1644.1375	76856.3096	.8102	.8053	1.6882	.7410	
	5.7623	.0007	1643.5902	74664.7910	.8147	.8098	1.6973	.7410	
	5.9610	.0004	1643.0432	72534.7910	.8190	.8141	1.7066	.7410	
	6.1597	.0002	1642.4963	70466.0264	.8233	.8184	1.7162	.7410	
	6.3584	.0001	1641.9496	68456.2656	.8276	.8226	1.7260	.7410	
	6.5571	.0000	1641.4031	66503.8252	.8317	.8268	1.7360	.7410	
	6.7558	.0000	1640.8568	64607.0703	.8357	.8308	1.7462	.7410	
	6.9545	.0000	1640.3108	62764.4126	.8396	.8348	1.7566	.7410	
	7.1532	.0000	1639.7647	60974.3076	.8433	.8387	1.7673	.7410	
	7.3519	.0000	1639.2187	59235.2617	.8468	.8425	1.7783	.7410	
	7.5506	.0000	1638.6733	57545.6135	.8501	.8463	1.7894	.7410	

\*\*\*\*OUTLET FROM MHD DUCT Point 3 of Fig. A 11.2.1

	GAS	LIQUID	TOTAL
TEMPERATURE, DEG. F	1178.44	1178.44	1178.44
PRESSURE, PSI	388.23	388.23	388.23
MASS FRACTION, REFERRED TO MHD DUCT INLET	.088771	.911229	1.000000
ENTHALPY, BTU PER POUND TOTAL AT MHD DUCT INLET	11.871	296.125	307.996
ENTROPY, BTU/R-LB(INLET MASS)	-.0005	.3165	.3161
AVAILABILITY, BTU/LB(INLET MASS)	12.1324	118.9562	131.0886
VOLUME FRACTION, LOCAL	.849981	.150019	1.000000
SPECIFIC VOLUME, CUBIC FEET PER POUND	1.1679	.0201	1.1220
MASS FLOW RATE, LB/SEC	727.89	7471.68	8199.57
VOLUME FLOW RATE, CU.FT/SEC	850.1	150.0	1000.1

\*\*\*\*INLET TO NOZZLE-SEPARATOR Point 4 of Fig. A 11.2.1

	GAS	LIQUID	TOTAL
TEMPERATURE, DEG. F	1178.44	1178.44	1178.44
PRESSURE, PSI	388.23	388.23	388.23
MASS FRACTION, REFERRED TO MHD DUCT INLET	.088771	.911229	1.000000
ENTHALPY, BTU PER POUND TOTAL AT MHD DUCT INLET	11.871	296.125	307.996
ENTROPY, BTU/R-LB(INLET MASS)	-.0005	.3165	.3161
AVAILABILITY, BTU/LB(INLET MASS)	12.1324	118.9562	131.0886
VOLUME FRACTION, LOCAL	.849981	.150019	1.000000
SPECIFIC VOLUME, CUBIC FEET PER POUND	1.1679	.0201	1.1220
MASS FLOW RATE, LB/SEC	727.89	7471.68	8199.57
VOLUME FLOW RATE, CU.FT/SEC	850.1	150.0	1000.1

\*\*\*\*GAS OUTLET FROM SEPARATOR Point 5 of Fig. A 11.2.1

	GAS	LIQUID	TOTAL
TEMPERATURE, DEG. F	1178.44	1178.44	1178.44
PRESSURE, PSI	388.46	388.46	388.46
MASS FRACTION, REFERRED TO MHD DUCT INLET	.088771	.000000	.088771
ENTHALPY, BTU PER POUND TOTAL AT MHD DUCT INLET	11.871	.000	11.871
ENTROPY, BTU/R-LB(INLET MASS)	-.0004	.0000	-.0004
AVAILABILITY, BTU/LB(INLET MASS)	12.0825	.0000	12.0825
VOLUME FRACTION, LOCAL	1.000000	.000000	1.000000
SPECIFIC VOLUME, CUBIC FEET PER POUND	1.1917	.0201	1.1917
MASS FLOW RATE, LB/SEC	727.89	.00	727.89
VOLUME FLOW RATE, CU.FT/SEC	867.4	.0	867.4

\*\*\*\*INLET TO REGENERATOR LOW PRESS SIDE\*\*  
Point 6/7 of Fig. A 11.2.1

	GAS	LIQUID	TOTAL
TEMPERATURE, DEG. F	150.00	150.00	150.00
PRESSURE, PSI	372.85	372.85	372.85
MASS FRACTION, REFERRED TO MHD DUCT INLET	.088771	.000000	.088771
ENTHALPY, BTU PER POUND TOTAL AT MHD DUCT INLET	.550	.000	.550
ENTROPY, BTU/R-LB(INLET MASS)	-.0112	.0000	-.0112
AVAILABILITY, BTU/LB(INLET MASS)	6.8010	.0000	6.8010
VOLUME FRACTION, LOCAL	1.000000	.000000	1.000000
SPECIFIC VOLUME, CUBIC FEET PER POUND	.4526	.0201	.4526
MASS FLOW RATE, LB/SEC	727.89	.00	727.89
VOLUME FLOW RATE, CU.FT/SEC	329.4	.0	329.4

\*\*\*\*INLET TO WASTE COOLER NUMBER 1 Point 8 of Fig. A 11.2.1

	GAS	LIQUID	TOTAL
TEMPERATURE, DEG. F	150.00	150.00	150.00
PRESSURE, PSI	365.40	365.40	365.40
MASS FRACTION, REFERRED TO MHD DUCT INLET	.088771	.000000	.088771
ENTHALPY, BTU PER POUND TOTAL AT MHD DUCT INLET	.550	.000	.550
ENTROPY, BTU/R-LB(INLET MASS)	-.0111	.0000	-.0111
AVAILABILITY, BTU/LB(INLET MASS)	6.7511	.0000	6.7511
VOLUME FRACTION, LOCAL	1.000000	.000000	1.000000

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SPECIFIC VOLUME, CU. FEET PER POUND	.4618	.0201	.4819
MASS FLOW RATE, LB/SEC	727.89	.00	727.89
VOLUME FLOW RATE, CU. FT/SEC	336.2	.0	336.2

\*\*\*INLET TO GAS COMPRESSOR NUMBER 1\*\*\*

Point 9 of Fig. A 11.2.1

	GAS	LIQUID	TOTAL
TEMPERATURE, DEG. F	120.00	120.00	120.00
PRESSURE, PSI	358.09	358.09	358.09
MASS FRACTION, REFERRED TO MHD DUCT INLET	.088771	.000000	.088771
ENTHALPY, BTU PER POUND TOTAL AT MHD DUCT INLET	.220	.000	.220
ENTROPY, BTU/R-LB(INLET MASS)	-.0115	.0000	-.0115
AVAILABILITY, BTU/LB(INLET MASS)	6.6818	.0000	6.6818
VOLUME FRACTION, LOCAL	1.000000	.000000	1.000000
SPECIFIC VOLUME, CU. FEET PER POUND	.4481	.0201	.4481
MASS FLOW RATE, LB/SEC	727.89	.00	727.89
VOLUME FLOW RATE, CU. FT/SEC	326.1	.0	326.1

\*\*\*OUTLET FROM GAS COMPRESSOR NUMBER 1\*\*\*

Point 10 of Fig. A 11.2.1

	GAS	LIQUID	TOTAL
TEMPERATURE, DEG. F	577.90	577.90	577.90
PRESSURE, PSI	1274.98	1274.98	1274.98
MASS FRACTION, REFERRED TO MHD DUCT INLET	.088771	.000000	.088771
ENTHALPY, BTU PER POUND TOTAL AT MHD DUCT INLET	5.261	.000	5.261
ENTROPY, BTU/R-LB(INLET MASS)	-.0107	.0000	-.0107
AVAILABILITY, BTU/LB(INLET MASS)	11.2739	.0000	11.2739
VOLUME FRACTION, LOCAL	1.000000	.000000	1.000000
SPECIFIC VOLUME, CU. FEET PER POUND	.2252	.0201	.2252
MASS FLOW RATE, LB/SEC	727.89	.00	727.89
VOLUME FLOW RATE, CU. FT/SEC	164.0	.0	164.0

COMPRESSOR 1 ENTROPIC EFFICIENCY = .8410

\*\*\*OUTLET FROM REGENERATOR HIGH PRESS SIDE\*\*\*

Point 11 of Fig. A 11.2.1

	GAS	LIQUID	TOTAL
TEMPERATURE, DEG. F	577.90	577.90	577.90
PRESSURE, PSI	1274.98	1274.98	1274.98
MASS FRACTION, REFERRED TO MHD DUCT INLET	.088771	.000000	.088771
ENTHALPY, BTU PER POUND TOTAL AT MHD DUCT INLET	5.261	.000	5.261
ENTROPY, BTU/R-LB(INLET MASS)	-.0107	.0000	-.0107
AVAILABILITY, BTU/LB(INLET MASS)	11.2739	.0000	11.2739
VOLUME FRACTION, LOCAL	1.000000	.000000	1.000000
SPECIFIC VOLUME, CU. FEET PER POUND	.2252	.0201	.2252
MASS FLOW RATE, LB/SEC	727.89	.00	727.89
VOLUME FLOW RATE, CU. FT/SEC	164.0	.0	164.0

\*\*\*LIQUID OUTLET FROM SEPARATOR \*\*\* Point 15 of Fig. A 11.2.1

	GAS	LIQUID	TOTAL
TEMPERATURE, DEG. F	1178.68	1178.68	1178.68
PRESSURE, PSI	368.81	368.81	368.81
MASS FRACTION, REFERRED TO MHD DUCT INLET	.000000	.911229	.911229
ENTHALPY, BTU PER POUND TOTAL AT MHD DUCT INLET	.000	296.125	296.125
ENTROPY, BTU/R-LB(INLET MASS)	.0000	.3166	.3166
AVAILABILITY, BTU/LB(INLET MASS)	.0000	118.9357	118.9357
VOLUME FRACTION, LOCAL	.000000	1.000000	1.000000
SPECIFIC VOLUME, CU. FEET PER POUND	1.2295	.0201	.0201
MASS FLOW RATE, LB/SEC	.00	7471.68	7471.68
VOLUME FLOW RATE, CU. FT/SEC	.0	150.0	150.0

\*\*\*INLET TO LIQUID PUMP\*\*\*

Point 16/17 of Fig. A 11.2.1

	GAS	LIQUID	TOTAL
TEMPERATURE, DEG. F	1178.68	1178.68	1178.68

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PRESSURE, PSI	368.81	368.81	368.81
MASS FRACTION, REFERRED TO MHD DUCT INLET	.000000	.911229	.911229
ENTHALPY, BTU PER POUND TOTAL AT MHD DUCT INLET	.000	296.125	296.125
ENTROPY, BTU/R-LB(INLET MASS)	.0000	.3166	.3166
AVAILABILITY, BTU/LB(INLET MASS)	.0000	118.9357	118.9357
VOLUME FRACTION, LOCAL	.000000	1.000000	1.000000
SPECIFIC VOLUME, CUBIC FEET PER POUND	1.2295	.0201	.0201
MASS FLOW RATE, LB/SEC	.00	7471.68	7471.68
VOLUME FLOW RATE, CU.FT/SEC	.0	150.0	150.0

\*\*\*DUCKET FROM LIQUID PUMP\*\* Point 18 of Fig. A 11.2.1

	GAS	LIQUID	TOTAL
TEMPERATURE, DEG.F	1185.60	1180.60	1180.60
PRESSURE, PSI	1250.00	1250.00	1250.00
MASS FRACTION, REFERRED TO MHD DUCT INLET	.000000	.911229	.911229
ENTHALPY, BTU PER POUND TOTAL AT MHD DUCT INLET	.000	299.632	299.632
ENTROPY, BTU/R-LB(INLET MASS)	.0000	.3167	.3167
AVAILABILITY, BTU/LB(INLET MASS)	.0000	122.2794	122.2794
VOLUME FRACTION, LOCAL	.000000	1.000000	1.000000
SPECIFIC VOLUME, CUBIC FEET PER POUND	.3632	.0201	.0201
MASS FLOW RATE, LB/SEC	.00	7471.68	7471.68
VOLUME FLOW RATE, CU.FT/SEC	.0	150.0	150.0

ENTHALPY ADDED AT TOP OF CYCLE = 11.98BTU/LB  
 ENTHALPY REJECTED AT BOTTOM OF MHD CYCLE = 11.65BTU/LB OF ILET MIXTURE  
 NET WORK = .33 BTU PER LB INLET  
 MHD DUCT ISENTROPIC EFFICIENCY = .7480  
 MHD TOPPING CYCLE EFFICIENCY = .0276  
 MAX. THERMODYNAMIC EFFIC. FOR ST. GEN. = .4621 FOR FEED WATER HEATER .0587 AND FOR PRECOOLERS .0000 .0000 .0000  
 STEAM BOIL. CYCLE EFFICIENCY = .3958  
 OVERALL COMB. CYCLE EFFICIENCY FOR IDEAL COMBUSTER-PRIMARY HT.X = .4071  
 OVERALL COMB. CYCLE EFFICIENCY INCLUDING COMBUSTER-PRIMARY HT.X AND PREHEAT ADVANTAGE = .3996  
 OVERALL COMB. CYCLE EFFICIENCY INCLUDING COMBUSTER-PRIMARY HT.X = .3574

TOP TEMPERATURES WITH VARIOUS SPLITS BETWEEN GAS AND LIQUID HEATING  
 FRACTION HEAT ADDED TO GAS \*\*GAS TEMP. F. LIQUID TEMP. F.

.0000	577.9015	1225.0498
.1000	686.7471	1220.6676
.2000	795.5926	1216.2841
.3000	904.4381	1211.9013
.4000	1013.2837	1207.5184
.5000	1122.1292	1203.1356
.6000	1230.9747	1198.7528
.7000	1339.8203	1194.3699
.8000	1448.6658	1189.9871
.9000	1557.5113	1185.6042
1.0000	1666.3569	1181.2214

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## Appendix A 11.3

### SUPERCONDUCTING MAGNET DESIGN FOR LIQUID-METAL MHD GENERATORS

#### A 11.3.1 General Magnet Requirement

The liquid-metal MHD generator is rather different from the open- and closed-cycle MHD generators. The major difference in the magnet design arises from the fact that the magnetic field in the duct is relatively small - below 2.0 T. The requirement for field uniformity is easily met by using iron poles to shape the magnetic fields. Hence, the base case magnet design was selected to have four ducts which were interconnected magnetically by iron, as previously shown in Figures 11.7 and 11.8 of the text. This method of construction reduced the stray fields, which is important from two aspects. The stray fields represent an environmental problem because magnetic objects tend to be pulled toward the magnet; and secondly, the stray field from one duct magnet will tend to affect the uniformity of the nearby magnets of the other MHD ducts. By minimizing the stray fields through the use of iron flux shunting paths, the above problems have been eliminated.

As with the open- and closed-cycle plasma MHD generators, the design of a superconducting magnet for LM-MHD application does not require any new technological developments with respect to the superconductor. Since the magnetic field is below 2 T at the wire, presently available multifilament niobium-titanium superconductors operating at 4.2°K (-452.13°F) were determined to be sufficient for this application.

Accordingly, the base case design herein described has been determined,utilizing the following design boundary conditions:

- A niobium-titanium filamentary conductor would be employed as the source of the magnetic field.
- The magnet winding would be forced cooled and operate at  $4.2^{\circ}\text{K} \pm 0.2^{\circ}\text{K}$  ( $-452.13^{\circ}\text{F} \pm 0.36^{\circ}\text{F}$ ).
- The magnet structure would be designed to be self-supporting against magnetic loading.
- The structure would be designed for an overall minimum cost.
- Iron poles between ducts would be used to shunt the magnetic field between the ducts and to provide a field uniformity within the duct to 8%.

#### A 11.3.2 Magnetic Field Analysis

The principal features of the electrical design for the base case LM-MHD generator are shown in Figures A 11.3.1 and A 11.3.2. Magnetic field calculations for this design were carried out using an existing computer program (References 11.15, 11.16, and 11.17) that is based on a numerical evaluation of a Fredholm-type integral for solving two-dimensional magnetostatics problems in rectangular coordinates. The uniformity of the field along the entire length of the generator ducts were determined by computing separately the fields for the generator entrance and exit cross sections.

The linear current sources were assumed constant and having a value of 1.1875 MA/m throughout the calculations. The total winding current required to achieve this is  $1.202 \times 10^7$  ampere-turns (At) at the generator entrance and  $2.649 \times 10^7$  At at the generator exit. The field calculations also assume that the iron pole width is 2.225 m (7.3 ft), that it is located 0.3048 m (1 ft) from the duct wall, and that it extends 0.3048 m (1 ft) beyond the duct corners.

Dwg. 6367A52

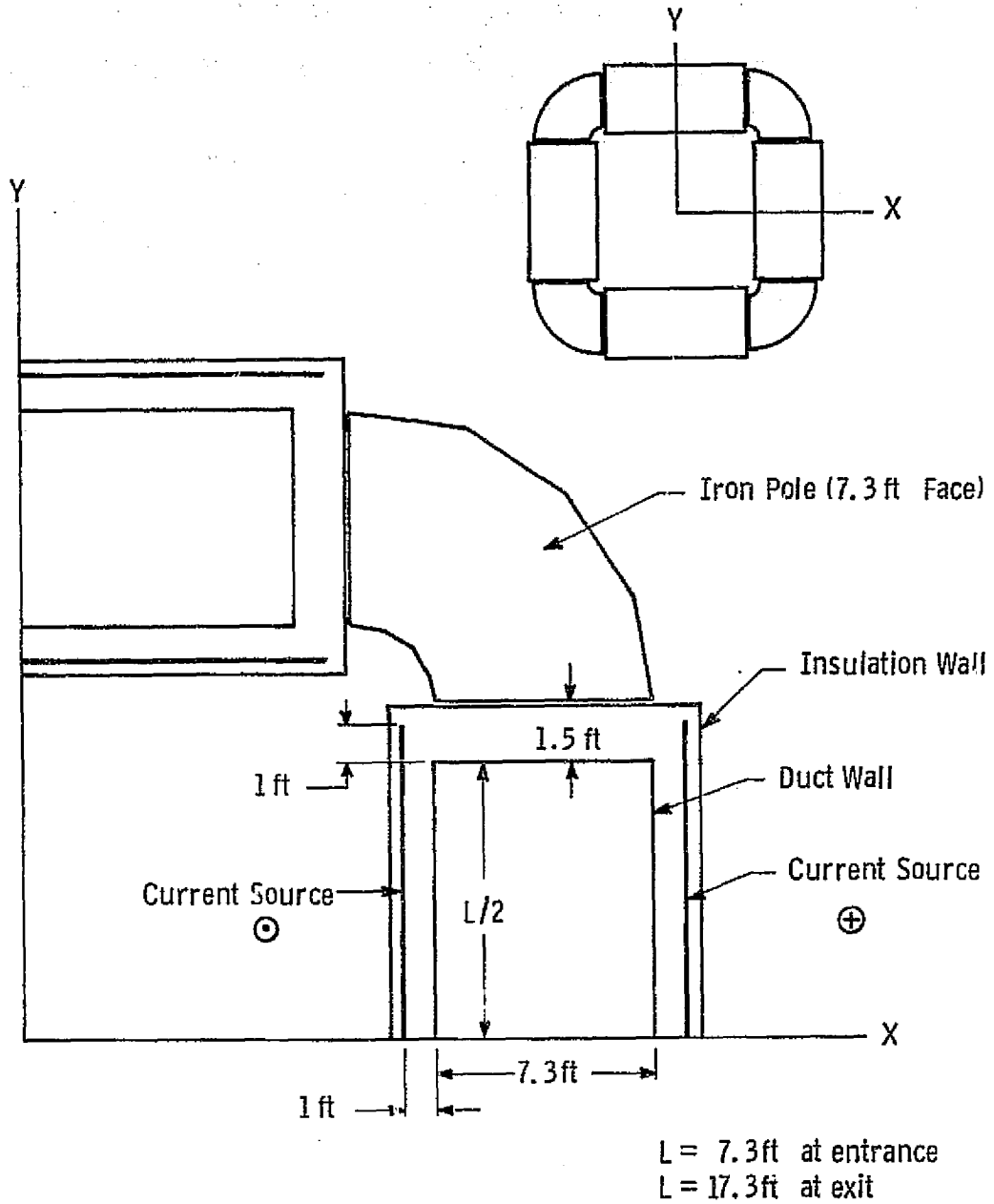
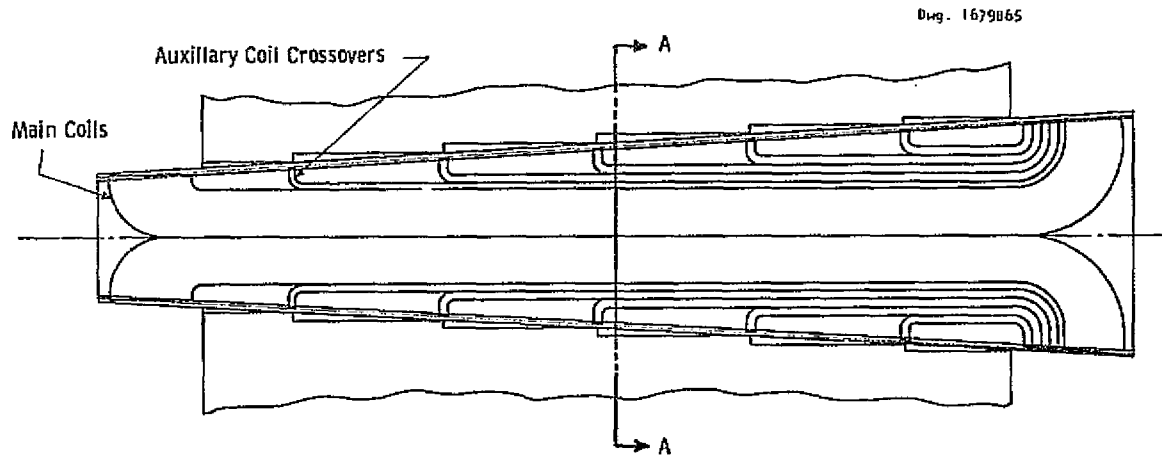


Fig. A 11.3.1 - The liquid-metal MHD generator magnet design employed in magnetic field calculations





Dwg. 1679065

Fig. A 11.3.2—Winding configuration for liquid metal MHD design

Curve 682867-A

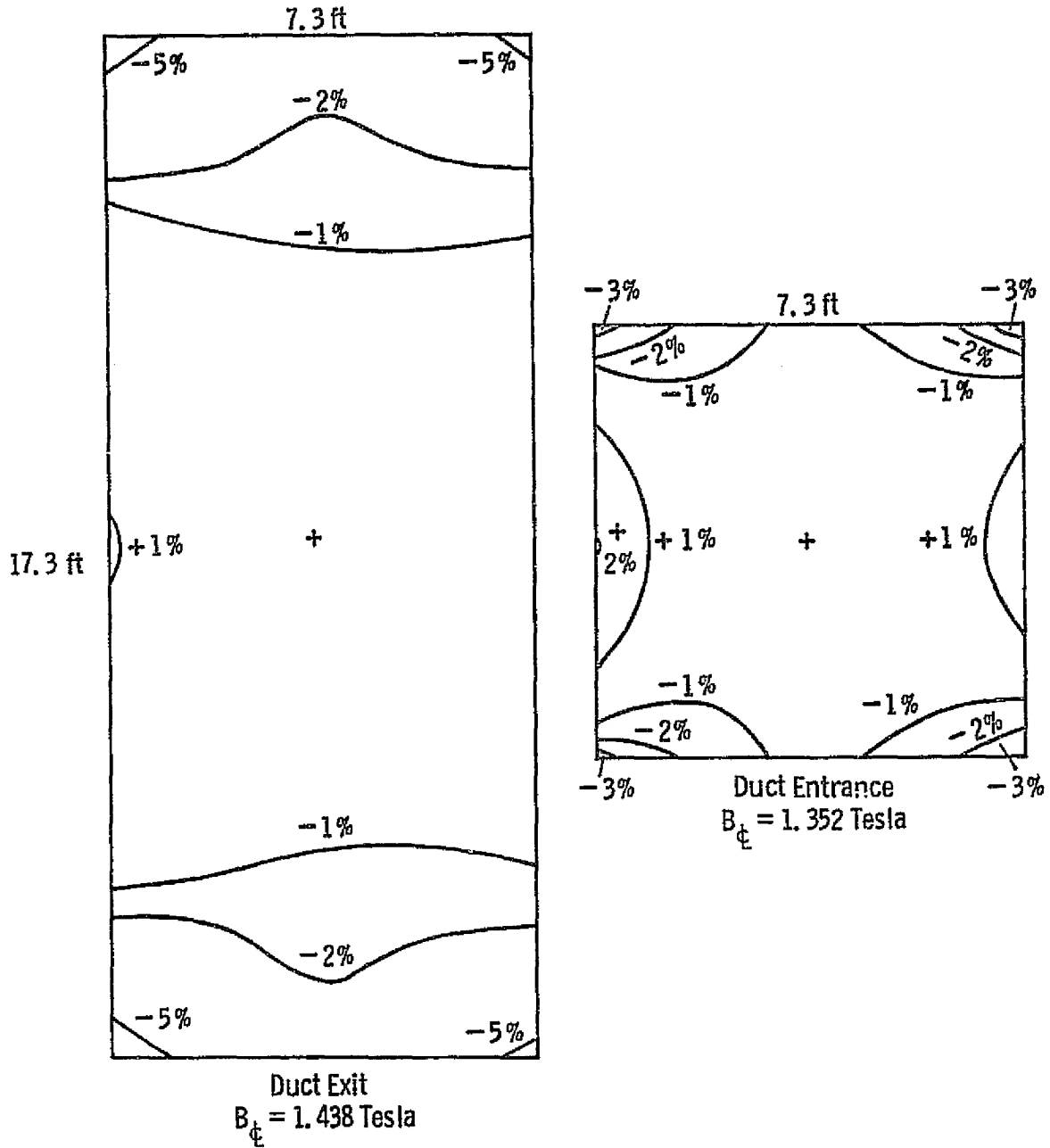


Fig. A11.3.3—Magnetic field uniformity for the exit and entrance faces of the liquid metal MHD generator duct

The calculated magnetic field uniformity across the entrance and exit duct faces is shown on Figure A 11.3.3, where lines of constant percent deviation from the value of the field at the duct centerline are given. The field uniformity is better than 4% at the entrance and 6% at the exit of the generator, where the centerline fields are 1.352 and 1.438 T, respectively. Based on the maximum and minimum fields in the generator ducts the overall field for the electrical design can be expressed as 1.373 T  $\pm$  5.6%.

If the iron pole width was increased from 2.225 to 2.53 m (7.3 to 8.3 ft) then the overall field was found to be 1.387 T  $\pm$  4.8%. This is an 0.8 point improvement in the spread of the field determined for the 2.225 m (7.3 ft) pole.

#### A 11.3.3 Conductor Design

The environment of a LM-MHD magnet system does not require the selection of sophisticated superconductors, primarily because there are no ac or transient magnetic fields present. Furthermore, the magnetic field distribution cannot accommodate graded winding designs because of the presence of cross-over turns linking the magnetic field.

Since the peak field seen by the duct for the LM-MHD concept is approximately 1.5 T or less, a filamentary niobium-titanium conductor has been selected for the base case design.

The most important aspect to the conductor design is selecting a reasonable operating current density for the winding. Illustrated in Figure A 11.3.4 is a normalized plot of the critical current density,  $j_c$ , for niobium-titanium wires as a function of the peak field on the wire ( $j_c(5T) \approx 10^9$  A/m<sup>2</sup>). An operating current of 5000 A was arbitrarily selected for this magnet. A conductor with filaments of 100  $\mu$ m (0.00394 in) or less, which is twisted approximately 39.4 twists per meter (1 twist per inch) was selected to inhibit the possibility of flux jump instability in the wire when the magnet is either being charged or discharged. A winding packing factor of 0.7 and a conductor aspect ratio of 2 to 1 were established as sufficient to accommodate liquid

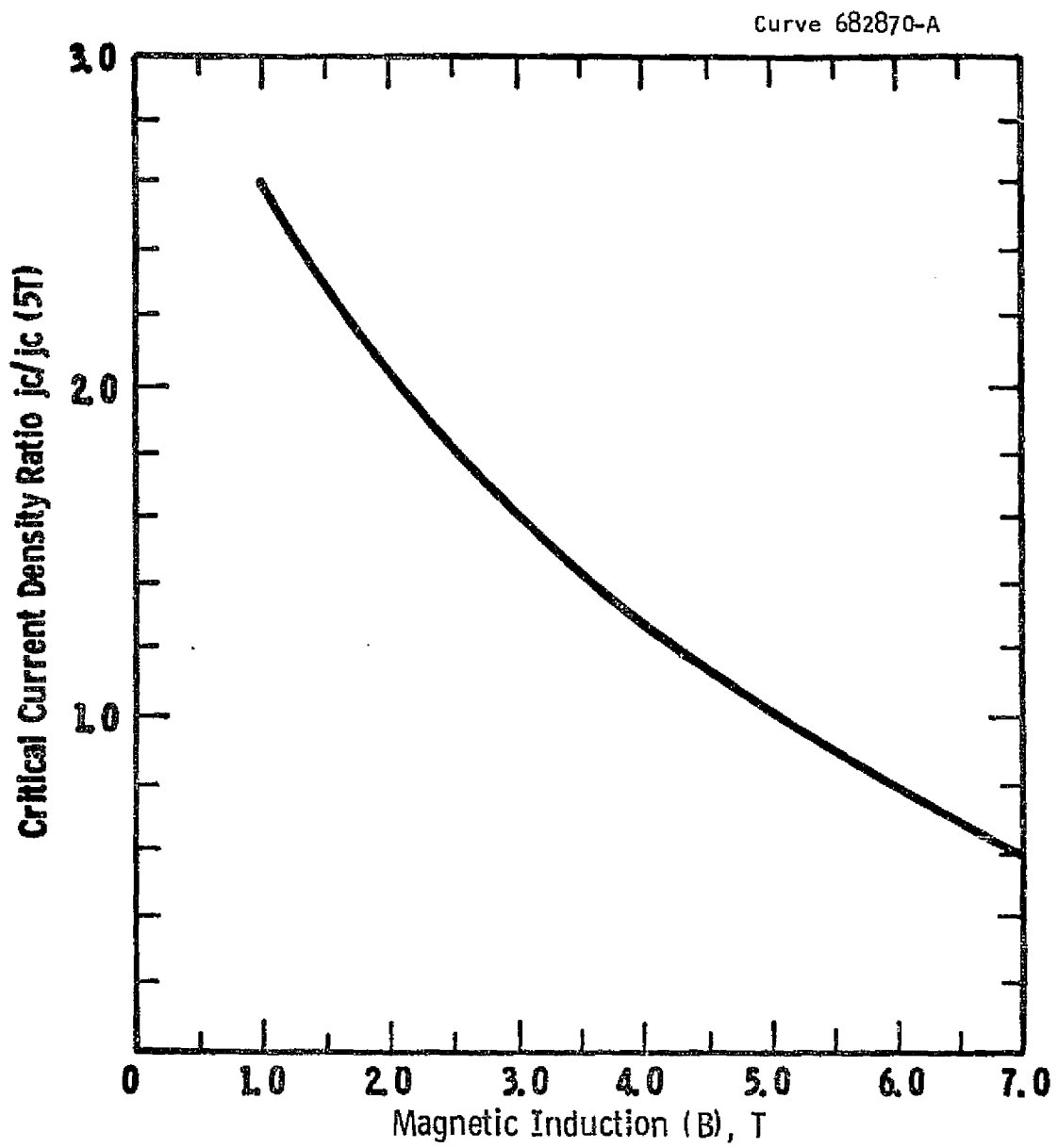


Fig. A 11.3.4—Normalized  $J_c$ -B curve for NbTi superconductors

helium cooling and the necessary distributed support structure. To ensure stable operation, a 3 to 1 copper-to-superconductor ratio was selected, which is not a sufficient amount of copper to crystallize the winding but should be sufficient to dynamically stabilize the winding. Although the MHD magnet is a dc device, provision for operation margin must be made. If one has temperature excursions of between 0.1 to 0.2°K (0.18 to 0.36°F) from the nominal 4.2°K (-452.13°F) in the windings, an operational current density  $j$  of about 0.5 jc at 4.2°K (-452.13°F) affords a reasonable compromise between high current density and thermal margin and was selected as the design point for the peak field region in the winding. Accordingly, a winding current density of  $2 \times 10^8$  A/m<sup>2</sup> was selected for the electrical design of the liquid-metal base case magnet design.

#### A 11.3.4 Stored Energy and Inductance Considerations

The stored energy of the system,  $E_s$ , can be calculated if we consider the iron to have infinite permeability in comparison to air from

$$E_s = \frac{B^2 V}{2\mu_0}$$

where  $B$  is the average magnetic interaction in the duct,  $\mu_0$  is the permeability of free space =  $4\pi \times 10^{-7}$  H/m and  $V$  is the total volume between the iron poles.

Since the typical magnets considered have large inductances, it is advised that the winding be subdivided and powered by separate power supplies to ensure reliable operation.

#### A 11.3.5 Mechanical Design

The major structural problem in the design of the dewars is the provision of a lightweight wall structure to minimize construction costs, that is compatible with the bending moments generated by the electromagnetic forces on the coils. Because of the necessity of

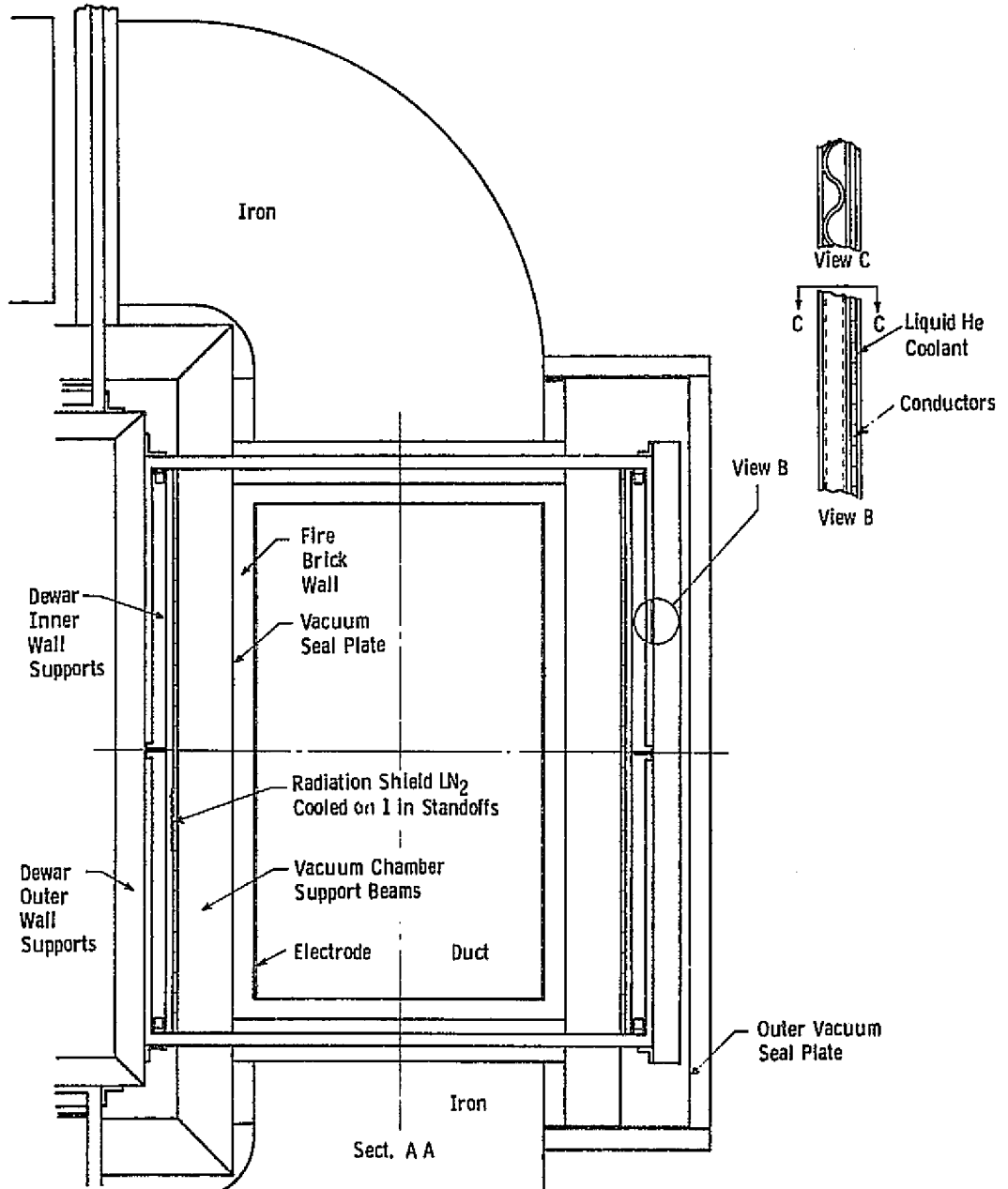


Fig. A 11.3.5-- Sectional view of liquid metal MHD magnet design

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enclosing liquid helium in the dewar, it is impractical from a heat transfer standpoint to provide supporting struts to the warmer structural elements. The dewars are to be essentially self-supporting, except for the provision of support columns at the base to sustain the dead weight of the structure.

For maximum economy and minimum weight the dewar walls subjected to bending loads were designed in plate-girder form. The spacing height and thickness of the webs were optimized with the thickness of the plates to produce a minimum weight structure consistent with the restriction of design stresses and buckling. In some cases, where other design considerations were dominant, a nonoptimal (in terms of weight) structure was used.

Figure A 11.3.5 shows the essential elements of the dewar design as discussed above.

#### A 11.3.6 Heat Transfer Analysis

The requirement of maintaining a superconducting magnet at 4.2°K (-452.13°F) within a tolerance of 0.2°K (0.36°F) can be achieved with existing technology. Liquid helium and nitrogen systems can be operated continuously for a year or more. In the past, the limiting factor in system endurance was clogging of the flow passages due to the freezing out of impurities. The major impurity source was compressor oil. Presently, dry compressors, tube expanders, and redundant compressors in a closed-loop, cryogenic refrigeration system can be operated continuously for several years. Cryogenic refrigeration equipment with sufficient reliability to support the superconducting magnets for the MHD generators, therefore, are available.

A survey of the manufacturers of refrigeration equipment was performed several years ago, and these results have been applied in estimating the electrical requirements of the refrigeration equipment. No allowance has been made for technological improvements that may be made to increase the percent of Carnot efficiency of these machines. It is expected, however, that the continued emphasis on utilizing

Table A 11.3.1

## Summary of Cooling Requirements for Liquid-Metal MHD Design

Load	Heat Input, W	Mass Flow, lb/hr	Electrical Load, kW
(a) Helium Refrigeration			
Radiation	18	20	10.8
Electrical Leads	100	113	180.0
Support Structure	40	45	24.0
Totals	158	178	214.8
(b) Nitrogen Refrigeration			
Helium Refrigerator	761	30.3	6.1
Radiation	7500	298.5	60.0
Conduction	325	12.9	2.6
Totals	8586	341.7	68.7



superconducting magnets in power generation systems will encourage innovation in the heat exchanger design and improvements in the reliability.

Liquid helium refrigeration systems can be obtained with specific power requirements (watts of electrical power required to produce one watt refrigeration) as low as 300 W/W and helium liquefaction systems with as low as 400 W/W. These types of specific powers are generally obtained only in very large capacity devices. For a total system load of 160 W, 600 W/W are required for liquefaction.

Liquid nitrogen refrigeration systems can be obtained with specific power requirements as low as 6.8 W/W. For a 9 kW load specific power of 8 W/W is required.

A cursory examination of the use of foam insulation around the periphery of the liquid nitrogen shield showed that the liquid nitrogen requirements for the radiation shield would be one-sixth of the requirements for the foam-insulated liquid nitrogen vacuum vessel.

Since the most economical system for a magnet operating in a power system is usually the one with the highest efficiency, the down loss shielding system was selected; although if accessibility is an important requirement, the foam insulation will not severely affect the overall efficiency.

Illustrated in Table A 11.3.1 is a summary of the cooling requirements for the LM-MHD magnet.

#### A 11.3.7 Summary of Magnet Cost Basis

The cost of the magnet system including refrigerators for LM-MHD magnets can be roughly estimated by multiplying the cost of the conductor by a factor of approximately four. Hence, the following parametric expression applies for a four-duct magnet system:

$$C = \frac{32 B L 10^6}{\mu_0 J_{av}} \left( \frac{D_1 + D_2}{2} + 2R' \right)$$

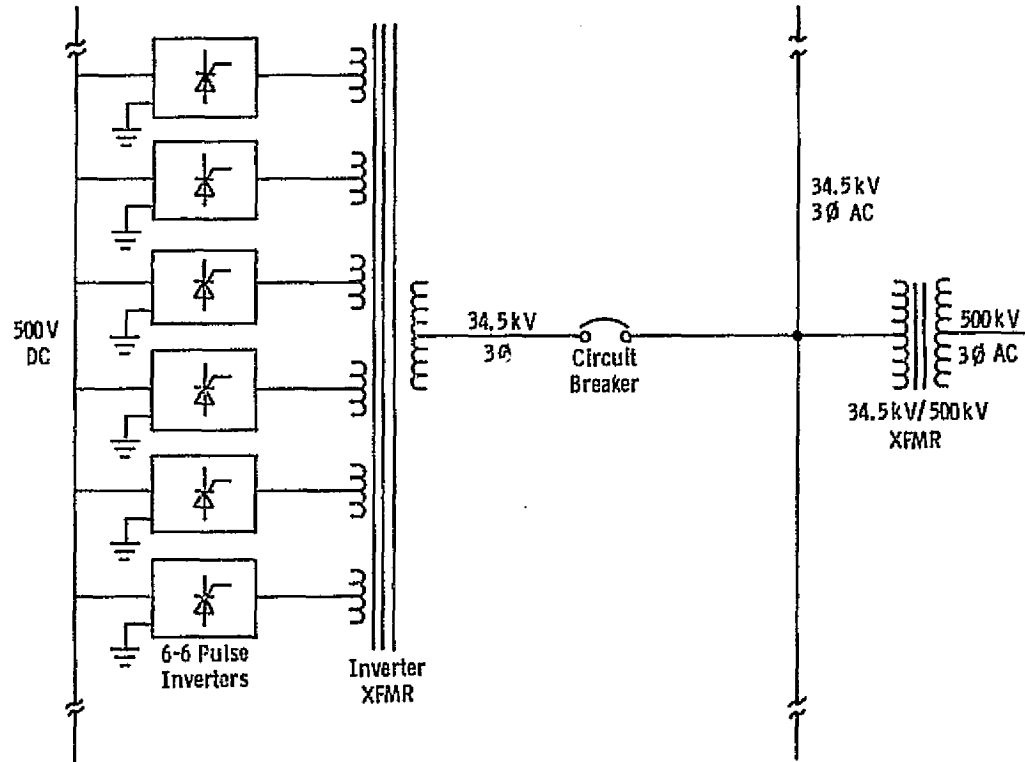
Table A 11.3.2 - Summary of the Liquid-Metal  
MHD Magnet Design

Nominal Rating	
Inlet Cross-sectional Area	2.2 m x 2.2 m
Exit Cross-sectional Area	5.2 m x 2.2 m
Length of Duct	22 m
Field on Axis of Duct	1.37 T
Number of Ducts	4
Average Ampere Turns Required	$1.93 \times 10^7$ At
Current per Turn	5000 A
Average Winding Current Density	$2 \times 10^8$ A/m <sup>2</sup>
Winding Peaking Factor	0.7
Conductor Aspect Ratio	2:1
Fraction of Superconductor in Conductor	0.25
Inductance	145 h
Stored Energy	1817 MJ
Number of Turns	5298
Conductor Operating Temperature	4.2°K
Liquid Helium Refrigerator Thermal Load	158 W
Liquid Helium Refrigerator Electrical Load	215 kW
Liquid Nitrogen Refrigerator Thermal Load	8.6 kW
Liquid Nitrogen Refrigerator Electrical Load	69 kW
Total Electrical Load	284 kW
Total Estimated Cost	\$15,400,000
\$/kVA	

where

- C = cost of magnet system, \$
- B = magnet field, T
- L = length of duct, m
- $\mu_0 = 4 \pi \times 10^{-7}$  H/m
- $J_{av}$  = average winding current density
- $D_1$  = inlet duct width, m
- $D_2$  = exit duct width, m
- $R'$  = insulation thickness, m.

Utilizing this information and the information presented earlier, a summary of the results for the base case liquid-metal design is illustrated in Table A 11.3.2. The unit cost is approximately \$116/kVA, and there are some economies with scaling to even larger magnet systems implicit to the cost expression presented.



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Fig. A 11.4. 1 - One-line diagram of dc to ac power conditioning circuit. One of 24 such 500 V d.c., - 34.5 kV converter circuit is shown

## Appendix A 11.4

### DC TO AC POWER CONDITIONING SYSTEM FOR LM MHD

#### A 11.4.1 General Requirements

A power conditioning system is needed to convert the dc power from the output of the liquid-metal MHD generator to ac power. the MHD electrical power output for this specific case is 1800 MW, of which 1000 MW is to be fed to the mains having a line-line voltage of 500 kV; and the remaining 800 MW will be used to drive the pumps and auxiliary equipment. The output voltage and current from the MHD generator are:

- Voltage - 500 V
- Current - 3.6 MA.

A one-line diagram of the power conditioning system is shown in Figure A 11.4.1. The dc power from the MHD busses is fed into 144 solid-state six-pulse inverter sets, each having a dc input voltage and current of 500 V and 25 kA. The output from each inverter set is fed to one of the six primary windings of the inverter transformer. It is proposed that 24 inverter transformers will be used. Each transformer will have six 350 V - 17.6 kA, three-phase primary windings; and a 34.5 kV, three-phase secondary winding. Fourteen of these inverter transformers will be connected to the primaries of the 500 kV power transformers. A 34.5 kV 2500 MVA, three-phase circuit breaker will be inserted between each inverter transformer and the power transformer. Three 500 kV, 333 MVA, single-phase power transformers are used in a three-phase connection. Power from output of the other ten inverter transformers will be employed to run pumps and other auxiliaries.

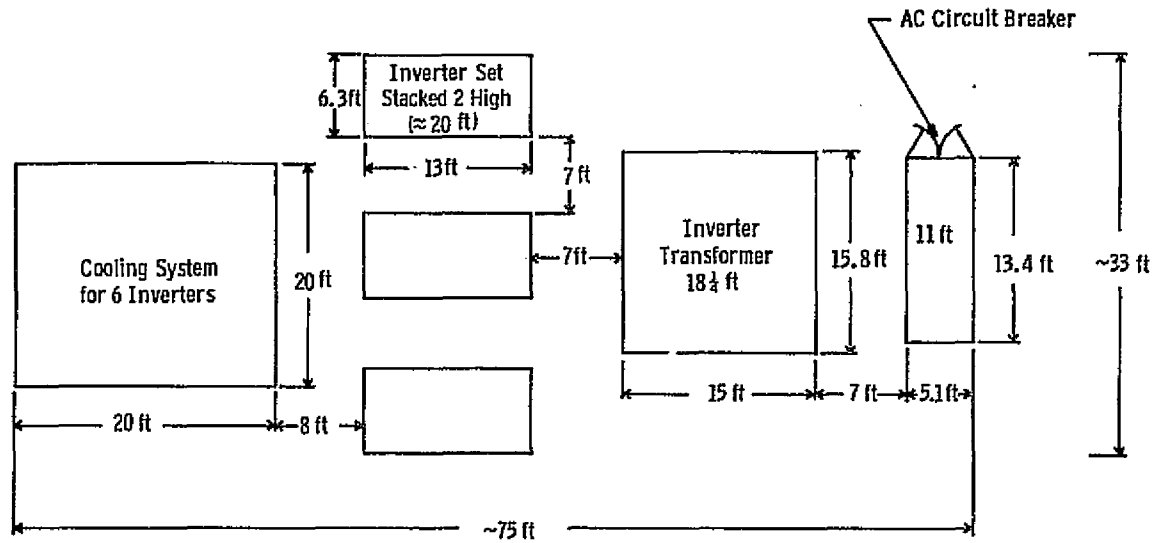


Fig. A 11.4.2—Layout of dc to ac converter circuit components.  
Twenty-four such circuits are arranged in parallel with each other

Each inverter set produces a three-phase output voltage with six current pulses. Since the inverter transformer has six three-phase primary windings, each primary receives six current pulses from the respective inverter. The inverters can be synchronized with each other so that the inverter transformer receives a total of 36 pulses. By proper timing of these inverters, the harmonic content of the output of the inverter transformer would be negligible. A filter across the output of the dc to ac converter system, therefore, is not necessary. This would represent a cost reduction of approximately \$6/kW to the converter system.

A 11.4.2 Size and Weights of the DC to AC Converter Components and Power Transformers

The dimensions and the weights of the inverter sets, inverter transformer, ac circuit breaker, and power transformer are given in Table A 11.4.1. The total weight of the system is 7159.4 Mg (7,890 tons). The dimensions of the pad necessary to house these components is approximately 36.57 by 304.8 m (120 by 1000 ft). Figure A 11.4.2 shows a suggested layout of one of the 24,500 V dc to 34.5 kV ac converter circuits.

A 11.4.3 Cost of Dc to Ac Converter Components

The cost of the components for the converter having an input power of 1800 MW dc and output power of 800 MVA at 34.5 kV and 1000 MVA at 500 kV are given as follows:

● 12.5 MW inverter set at \$30/kW is	\$ 375,000
For 144 inverter sets is	54,000,000
● 83 MVA 350 V/34.5 kV inverter transformer is	470,000
For 24 inverter transformer is	11,280,000
● 34.5 kV, 2000 A, 2500 MVA circuit breaker is	24,200
For 24 circuit breakers is	581,000
● 34.5 kV/500 kV, 333 MVA power transformer is	832,000
For three power transformers is	2,497,000

The total cost at FOB is \$68,358,000 which represents a cost of \$37.9/kW. These costs do not include delivery and installation.

Table A 11.4.1 - Size and Weights of DC and AC Power Conditioning Components

Item	Dimensions, in	Weight, lb
Inverter Set (144)	120 x 151 x 75.6	40,600
Inverter Transformer (24)	219 x 180 x 190	180,800
ac Circuit Breaker (24)	131.7 x 61.0 x 160.4	9,620
Cooling System for 6 Inverters (24)	146 x 240 x 240	240,000
Power Transformer (3)	465 x 260 x 210	445,000
Total Weight		15,783,500



## Appendix A 11.5

### LIQUID-METAL SYSTEM AND SUBSYSTEMS

#### A 11.5.1 General Requirements

The liquid-metal system and subsystems consist of the following major items:

- Liquid-metal pump
- Liquid-metal piping
- Liquid-metal purification
- Liquid-metal dump systems.

Much of the liquid-metal (sodium) technology that exists today has been developed in conjunction with the liquid-metal fast breeder reactor (LMFBR). Current sodium technology is reflected in both the fast flux test facility (FFTF) and Clinch River Breeder Reactor Program (CRBRP) demonstration plant. Table A 11.5.1 summarizes the key design parameters.

#### A 11.5.2 Liquid-Metal Mechanical Pumps

The basis for determining the MHD liquid-metal pump cost was current state-of-the-art design practice for the breeder reactor programs. The sodium pumps presently being considered for the CRBRP are on the order of  $1.8927 \text{ m}^3/\text{s}$  (30,000 gpm) each and based on single-stage, low-head [1.063 MPa (450 ft) sodium] hydraulic design. Figure A 11.5.1 shows a schematic diagram of one such pump design. Costs for this type of sodium pump are estimated at \$3.4 million per pump.

The liquid-metal pump for the MHD system under study requires pumps to handle flows in the range of 15.77 to  $31.545 \text{ m}^3/\text{s}$  (250,000 to

Table A 11.5.1 - Construction Seismic Class 1

Liquid-Metal System	FFTF	Clinch River
LM pump	Mechanical, single-stage 15,000 gpm, 450 ft Na	Mechanical, single-stage 30,000 gpm, 450 ft Na
Pumping and Valves	Piping and valving 28-36 in diameter T = 1200°F V = 20-40 ft/s p = 200 psi	Piping and valving 36-50 in diameter T = 1200°F V = 20-40 ft/s p = 200 psi
Purification (Na)	2 ppm O <sub>2</sub> level cold trap O <sub>2</sub> , H <sub>2</sub> monitoring 0.1% of total volume per day; cover gas leak rate (max)	2 ppm O <sub>2</sub> level cold trap O <sub>2</sub> , H <sub>2</sub> monitoring
Safety-Emergency	Sodium-dump - gravity smoke and sodium vapor leak detectors	Sodium dump

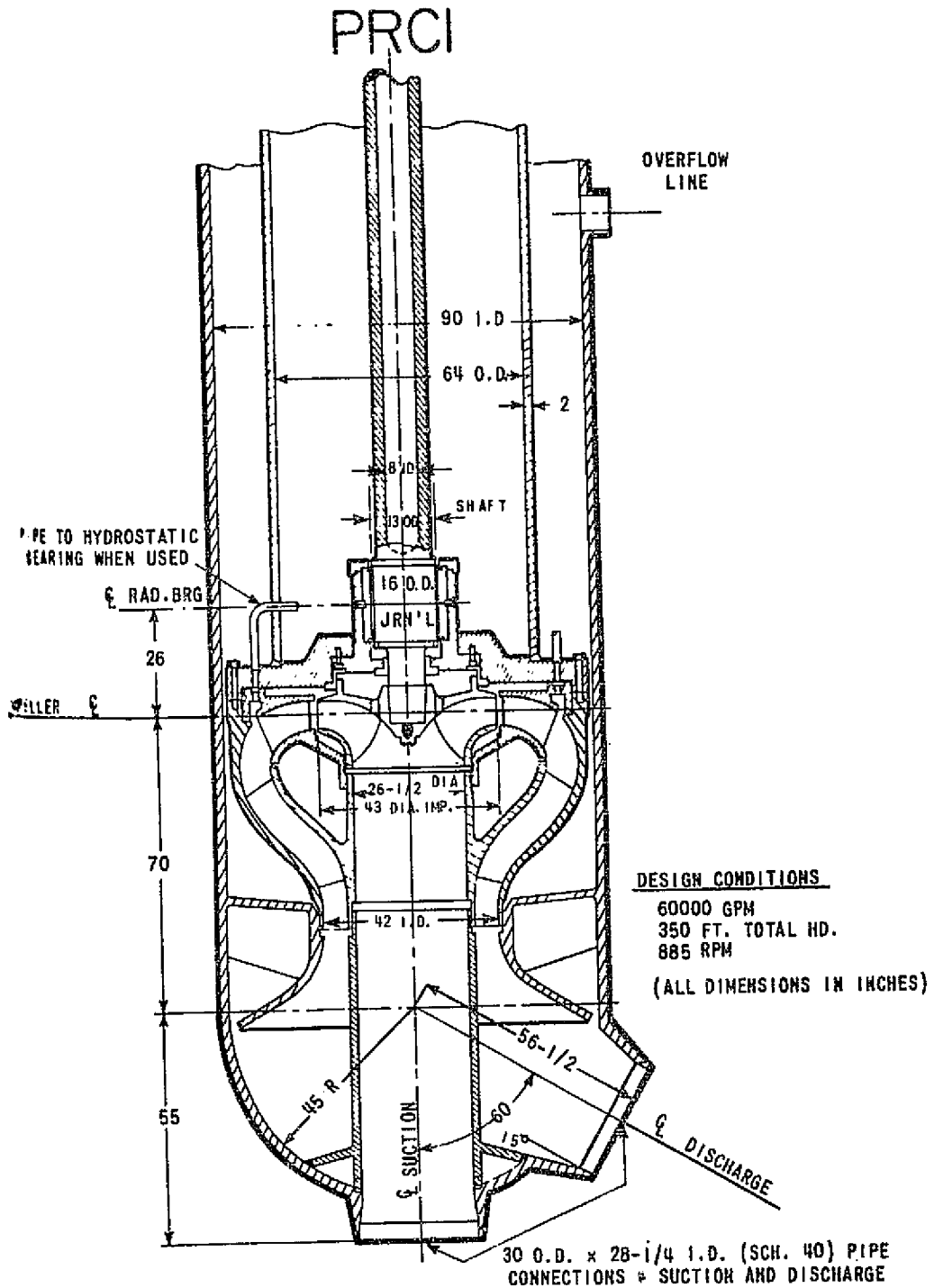


Fig. A 11.5.1—Single-suction pump (Removable concentric casing)

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500,000 gpm) at high system heads [5.472 MPa (54 atm)]. A preliminary design of a sodium pump to meet these requirements was prepared by the Westinghouse Cheswick Large Pump Division. The design was based on current sodium pump technology as embodied in the CRBRP design shown previously in Figure A 11.5.1. The basic and essential hydraulic features of the LM-MHD sodium pump design are summarized in Table A 11.5.2.

The cost for this pump was developed on the basis of material requirements. This cost, however, was subsequently judged to be excessive, and substantial reduction in the material usage could be achieved (particularly with the casing). Such a design effort, however, was beyond the scope of work intended for this study program.

It was ultimately concluded that a reasonable pump cost estimate could be made because of the similarity of the hydraulic designs. Thus, it was decided to extrapolate the Clinch River pump costs to the present LM-MHD case based on the number of stages and flow capacity. It is anticipated that the costs for higher LM-MHD pressure design conditions are somewhat offset by the costs for the Clinch River pump's nuclear code construction requirements.

The large liquid-metal flows and high-pressure heads require enormous pump power requirements 1020 MW shaft ( $1.37 \times 10^5$  hp). For the four-loop system, the specific drive requirements are 25.5 MW shaft (34,250 hp) per pump. With such large power requirements, direct coupling with a steam turbine drive from the steam plant should be considered. Two of the pumps could be driven in this manner. The remaining two pumps require motor drives.

For Point 11, the liquid-metal mechanical pumps are replaced with EM pumps. A preliminary design of an EM pump was prepared in order to determine size and cost of the system. Table A 11.5.3 summarizes the pump design data. Costing of this pump was based on weight at 517.6/kg (\$8/lb).

Table A 11.5.2 - Summary of Basis for Design  
of LM-MHD Sodium Pump

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DESIGN BASIS

Flow	-	250,000 gpm
Temperature	-	1200°F
Required Head	-	2320 ft Na
Sys. Operating Temp.	-	1200°F
Thermal Transients	-	None

PUMP DESIGN FEATURES

No. of Stages	-	5
Head/Stage	-	465 ft
Speed	-	500 rpm
Hydraulic Eff.	-	85%
BHP	-	146,000 hp
Impeller Dia.	-	80 in
Diffuser Dia.	-	120 in
Cross-over id	-	158 in
Casing id	-	195 in
Impeller Eye	-	53 in (suction end)
Shaft Dia.	-	26 in
Tank Wall	-	12 in

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Table A 11.5.3

SC Electric Machinery Systems High Field dc Conduction Pump

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Approximate Data - 4 units as follows:

Flow	- 250,000 gpm, 1200°F, 2300 ft head
Weight	- 380,000 lb/unit
Size	- MHD channel to pump:
	Transition section - 10 ft long
Pump inlet	- 6 ft long by 30 ft od
Pump	- 9 ft long by 30 ft od
Pump exit	- 6 ft long by 30 ft od
Pump efficiency	- 90%
Power input	- 95.303 per unit, 1.3236 MA at 72 volt

---

A 11.5.3 Piping and Valves

Piping for the liquid-metal flows would be mostly designed and costed in a manner similar to that of the MHD ducts, in other words, utilizing steel plates and rib supports. Circular piping might be required at the manifolding to the primary heat exchanger and perhaps at the pump connections. The design for the piping system sizing is based on 6.094 m/s (20 to 40 ft/s) velocity, compatible with present breeder reactor technology. Design of the piping system would need to include provision for thermal expansion, either by use of expansion loops or by designing equipment interfaces with slip-joint connections. The piping system would be constructed from 316 SS and designed for 946.08 Ms (30 yr) life. At the large piping size requirements, valving of the main liquid-metal flow is not anticipated.

The algorithm derived for costing and sizing the liquid-metal ducting was previously given in Table 11.4.

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The gas stream piping was divided into hot and cold legs. The respective length of each leg was estimated from the preliminary plans and elevation equipment arrangement sketches of the plants. Values for the length of hot and cold leg piping were approximated at 30.48 and 38.1 m (100 and 125 ft), respectively. The algorithms for sizing and costing the gas piping were given in Table 11.4. The calculations were made assuming a 30.48 m/s (100 ft/s) gas velocity. The equation is derived from:

- Mass continuity for the gas
- Equation for weight of circular pipe
- Stress equation for circular pipes.

These equations were combined to eliminate pipe diameter and thickness from the final expression. The cost coefficient in the weight equation was based on current design estimates for CRBRP large seamless stainless steel pipe.

#### A 11.5.4 Liquid-Metal Purification Subsystem

The liquid-metal purification system is required to maintain the concentration of certain contaminants at or below specified values. One of the most important of these contaminants is oxygen. High oxygen levels coupled with high temperatures can lead to severe corrosion of metal containment pipes and ducting. The corrosion of heat exchange tubes leads to reduced performance and ultimately the possibility of tube failure. Section 3.8 discusses specific materials requirements for sodium and lithium MHD systems.

With the liquid-metal (sodium) MHD system under study, the required sodium purity level (and hence, purification system requirements) may not be as stringent as those encountered in the breeder reactor designs. This observation is based on two factors: (1) liquid-metal temperature gradients in the MHD system are not severe (minimize mass transfer effects), and (2) high-pressure construction (thick walls) could relax corrosion allowances. Fouling of heat exchanger surfaces, however,

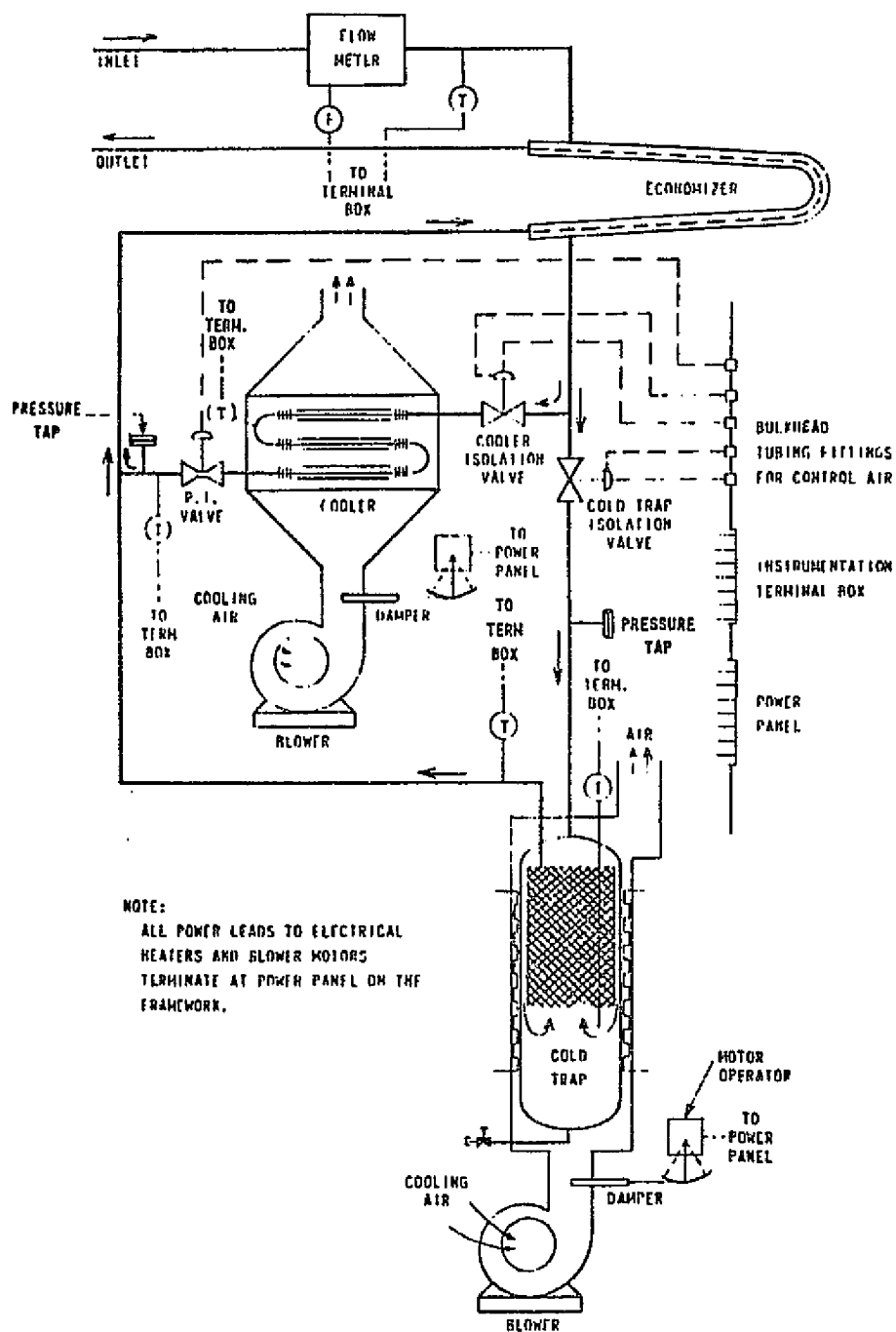


Fig. A 11.5.2—Flow diagram - Oxide control and indicating system



compatibility of the liquid metal with the MHD duct materials, and plugging of flow passages with crud must still be considered in specifying purity requirements.

The liquid-metal MHD system cold trapping will require reducing sodium temperatures from above 922 to 394°K (1200 to 250°F). A schematic of the cold trap system and components is shown in Figure A 11.5.2. The system consists of a pump, flowmeter, regenerator, heat sink, cold trap tank, and associated instrumentation. The instrumentation would function to monitor the performance and effectiveness of the cold trap system to remove sodium impurities, and should include oxygen and hydrogen meters both before and after the cold trap and a  $\Delta p$  gage across the cold trap to monitor the pressure drop associated with impurity build-up in the cold trap mesh.

The design (and subsequent costing) of the LM-MHD purification system has been based on the following assumptions:

- The diffusion of oxygen and hydrogen through piping and duct walls is negligible.
- The oxygen, hydrogen, carbon dioxide, and water vapor concentrations in the sodium will occur via gettering from the argon gas.
- Argon purity as received (prepurified grade) will be:<sup>\*</sup>

4 to 5 ppm (by volume)	O <sub>2</sub>
1 ppm	CO <sub>2</sub>
5 ppm	H <sub>2</sub>
7 ppm	N <sub>2</sub>
< 1 ppm	total hydrocarbons
5 ppm	H <sub>2</sub> O
- Argon makeup requirements will be based on a leak rate per day of 0.1% of total volume.

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\* Private communication with representative from Matheson Gas Products.

An equation that relates the required purification flow rate to the purification requirements is given as:

$$y = Z \left[ 1 - \frac{N}{N + \left( \frac{\rho_l}{\rho} \right) \frac{(s)(c)}{(MR)}} \right] \quad (\text{A } 11.5.1)$$

where

- y = required liquid-metal purification flow, gpm
- N = allowed oxygen concentration on the liquid metal, ppm
- $\rho$  = density, g = gas, l = liquid
- s = argon leak rate
- c = concentration of oxygen in the argon gas makeup, ppm
- MR = mass flow rate ratio in MHD loop, i.e., liquid-metal gas
- Z = total MHD liquid-metal flow, gpm.

For the specific conditions of sodium/argon with 0.1% leak rate under base case conditions, the equation reduces to:

$$y = Z \left( 1 - \frac{N}{N + 1} \right) \quad (\text{A } 11.5.2)$$

The cost model for the purification system was taken as:

$$C, \$ = 8572(y)^{0.6}$$

where C is the cost of the purification system in dollars and y the required capacity from Equation A 11.5.1. The constant 8572 and exponent 0.6 were determined on the basis of two cost estimates obtained from MSA

Research Corporation, supplier of liquid-metal oxide control systems. One of the cost quotations was based on a state-of-the-art 60 gpm flow rate, and the second was an estimate for an advanced state-of-the-art, 500 gpm purification system.

#### A 11.5.5 Liquid-Metal Inventory and Emergency Dump

The LM-MHD dump system was sized and costed on the basis of a very similar system now being designed for the CRBRP. This system consists of tanks, dump heat exchangers, valves, piping, pumps, and appropriate heat tracing. The tanks are lined carbon steel with capacities ranging up to  $189.3 \text{ m}^3$  (50,000 gal). Drain lines range up to 45.72 cm (18 in) diameter. Costs for this system for CRBRP are expected to be about \$4 million for  $302.8 \text{ m}^3$  (80,000 gal) capacity. This does not include sodium inventory costs. A temperature capability of up to  $700^\circ\text{K}$  ( $800^\circ\text{F}$ ) is maintained for each tank to prevent metal solidification.

The costing for the liquid-metal dump system for the LM-MHD plant was based on the above described system using a 0.7 power scaling law. The required liquid-metal capacity was determined from piping and component volume calculations.

Sodium inventory costs were determined on the basis of the required capacity and costed at \$3.01/kg (\$1.37/lb). Present production capacity is limited; this price, therefore, includes the cost of storing and handling the large quantities of sodium required for the LM-MHD system and is thus higher than that usually charged for small quantities [ $< 454 \text{ kg}$  ( $< 1000 \text{ lb}$ )]. If production capacity increases, the cost of sodium can be expected to decrease, but in any case the impact of a lower price on total plant costs would be insignificant.

Figure A 11.9 illustrates the relative positioning of the dump tanks with respect to the MHD duct assembly.

## Appendix A 11.6

### COUPLING HEAT EXCHANGERS

#### A 11.6.1 General Methodology for Coupling Heat Exchangers

When the LM-MHD coupling heat exchangers were designed, the general theoretical methodology followed that previously described in Appendices A 9.2 and A 9.3 of the open-cycle MHD study. The general correlations and equations for calculating the various heat transfer coefficients are summarized there. In adapting this analysis to the present study, some simplification and alternative approaches were made. These are summarized here, along with the final results. Simplified size and cost algorithms were developed for the following heat exchanger systems:

- Primary (fired) coupling heat exchanger
- MHD-steam plant coupling heat exchanger
- Air-gas preheaters (or precoolers)
- Gas recuperators.

#### A 11.6.2 Primary Heat Exchanger Design Basis

In order to determine a realistic costing for the primary heat exchanger for the closed-cycle LM-MHD cycles, a series of heat exchanger calculations was performed for sodium-, argon-, and steam-cooled heat transfer surfaces. In general, the design practice of state-of-the-art cyclone-furnace-fired, water-cooled boilers was used for guidance on tube geometry, overall dimensions, and materials of construction.

No attempt was made in any of the calculations to optimize the designs or to produce parametric design data. Rather, the calculations were used to provide rough engineering estimates of the cost of these types of heat exchangers.

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For the radiation section, sodium- and/or steam-cooled tubes were selected. For the 22.9 to 30.5 m (75 to 100 ft) high chambers, 5.08 cm (2 in) id tubes were selected for the calculations. Haynes alloy No 188 was chosen for these tubes because of its high-temperature creep strength. Also, these tubes were assumed to be clad (flue-gas side) either with one of the stainless steels or with some other oxidation-resistant coating. The usual wall thickness required for both strength and corrosion was 0.635 cm (0.25 in). For radiation exchange calculations, the 6.35 cm (2.5 in) od tubes were spaced on 12.7 cm (5 in) centers and located in a plane 13.97 cm (5.5 in) inside the refractory walls.

Once the hot products of combustion were cooled from the maximum flame temperature of around 2444°K (3940°F) in this radiation section, a second argon-cooled tube bundle was employed to further reduce the temperature of the combustion gases (while warming the argon to its final temperature). Once again, in this section of the heat exchanger the superalloy Haynes alloy No 188 was selected because of its high-temperature creep strength.

This second radiation-convection section of the exchanger was generally the most expensive section (on a \$/Btu/hr of heat transferred basis) because the argon-cooling heat exchange coefficients were always 5 to 30 times lower than those for steam or sodium, respectively. Also, the tube external heat transfer coefficient in this section is about 4 times lower than in the radiation section. Finally, the tube wall temperatures are high enough here to require a substantial tube wall constructed from a high-cost superalloy.

The final section of the exchanger was designed to cool the flue gases to the desired 644°K (700°F) exit temperature. In this convection section, finned tubes were specified in the tube bundles to enhance the external heat transfer and to reduce the number and length of the tubes.

Tables A 11.6.1 and A 11.6.2 summarize the most significant physical characteristics and heat transfer parameters of the design for the base case and for parametric Point 8.

In order to scale the cost and size of the primary heat exchanger for the various parametric cases, a simplified analytical model was developed for estimating heat transfer surface areas and related parameters. The analytical model and calculation procedures were based on the concept of the Number of Heat Transfer Units (NTU), a term introduced by London and Kays (Reference 11.10). The physical model of the primary heat exchanger was simplified from the real case to consider only a radiation section followed by a straight convection section as depicted in Figure A 11.6.1.

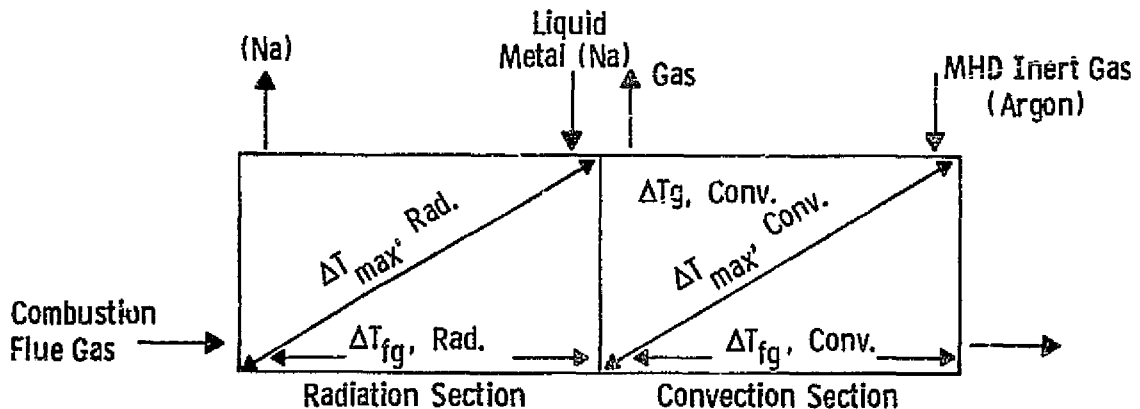


Fig. A 11.6.1 - Primary heat exchanger models

The heat capacitance ( $\dot{m} C_p$ ) of the liquid metal was large when compared to that of the combustion gases, and its thermal resistance negligible. The required primary heat exchanger heat transfer surface areas were determined from:

$$A_{T_{conv}} = (NTU)_{conv} C \left( \frac{1}{U} \right) \quad (A 11.6.1)$$

where  $C = (\dot{m} C_p)_{\text{argon}} (\Delta T_g / \Delta T_{fg})_{\text{conv}}$ ,  $U$  is the overall convective heat transfer coefficient (see Section 9, Appendix A 9.2), and:

$$A_{T \text{ RAD}} = (\text{NTU})_{\text{RAD}} C_{fg} \left( \frac{1}{h_{\text{RAD}}} \right) \quad (\text{A } 11.6.2)$$

where  $C_{fg}$  is the specific heat of the combustion gas and  $h_{\text{RAD}}$  (see Section 9, Appendix A 9.2) is the radiation heat transfer coefficient.

For the various parametric cases, then, the required total heat exchanger tubing length,  $L_t$ , was scaled from the appropriate case previously summarized in either Table A 11.6.1 or A 11.6.2 (low- or high-system temperature, respectively) according to their heat transfer surface area ratio (this assumes the same tube geometry).

The characteristic dimensions of the heat exchanger (diameter and length) were determined on the basis of the required gas and combustion products flow areas, required heat transfer surface area, and specified pressure drop. The equations for the characteristic diameter and length were previously listed in Table 11.5 of the text. The expression for diameter was developed on the basis of total flow cross-sectional area ( $A_{\text{flue gas}} + A_{\text{inert gas}} = \frac{\pi}{4} d^2$ ) and the known mass ratios between the two flows ( $\dot{m}_{\text{flue gas}} / \dot{m}_{\text{inert gas}} = 0.190$ ). For this calculation, the mass ratio was assumed to be constant for all cases. The numerical constant 28 appearing in Table 11.5 represents four times the mass ratio times the ratio of the product of velocity and density of the inert gas and combustion gas, respectively. This relation is derived from continuity considerations. It was further assumed that the flue gas velocity was approximately twice the inert gas velocity. Typical flue gas velocities ranged, therefore, from 15.24 to 30.48 m/s (50 to 100 ft/s).

The gas flow cross-sectional area,  $A_c$ , was calculated from the pressure drop and mass continuity considerations. The equation is:

Table A 11.6.1 Summary of Primary Heat Exchanger Design,  
Base Case

Item	Specifications
1	Combustor flue gas $8.69 \times 10^6$ lb/hr
2	Total heat load $8.5 \times 10^9$ Btu/hr
3	Liquid sodium flow $4.38 \times 10^8$ lb/hr
4	Sodium temperature rise 1181.4 - 1200°F ( $\Delta T = 18.6^\circ F$ )
5	Argon flow $4.27 \times 10^7$ lb/hr
6	Argon pressure 1250 psi
7	Argon temperature rise 560 - 1200°F ( $\Delta T = 640^\circ F$ )
8	Primary steam flow $5.124 \times 10^6$ lb/hr at 3500 psi
9	Primary steam temperature rise 800 - 1000°F ( $\Delta T = 200^\circ F$ )
10	Reheat steam flow $5.124 \times 10^6$ lb/hr at 600 psi
11	Reheat steam temperature rise 550 - 1000°F ( $\Delta T = 450^\circ F$ )

Design Data - Radiation Section (steam cooled)

Physical

1. Tube id	0.9 in
2. Tube wall thickness	0.285 in
3. Tube material	Haynes Alloy No. 188 - clad with oxidation resistant material
4. Tubing required	34,355 ft

Thermal

1. Radiation heat transfer coefficient	$50 \text{ Btu/hr-ft}^2\text{-}^\circ F$
2. Steam side heat transfer coefficient	$550 \text{ Btu/hr-ft}^2\text{-}^\circ F$
3. LMTD	1929°F
4. Heat load	$1.123 \times 10^6$ Btu/hr



Table A 11.6.1 Continued

Item	Specifications
Design Data - Radiation Section (sodium cooled)	
<u>Physical</u>	
1. Tube id	2.0 in
2. Tube wall thickness	0.25 in
3. Tube material	Haynes Alloy No. 188 - clad with oxidation resistant coating or metal
4. Tubing required	34,812 ft
<u>Thermal</u>	
1. Radiation heat transfer coefficient	55 Btu/hr-ft <sup>2</sup> -°F
2. LMTD	1970°F
3. Heat load	2.54 x 10 <sup>9</sup> Btu/hr
Design Data - Radiation/Convection Section (argon cooled)	
<u>Physical</u>	
1. Tube id	2.0 in
2. Tube wall thickness	0.25 in
3. Tube material	304 SS
4. Tubing required	99,695 ft
<u>Thermal</u>	
1. Heat transfer coefficient - external	15 Btu/hr-ft <sup>2</sup> -°F
2. Heat transfer coefficient - internal	100 Btu/hr-ft <sup>2</sup> -°F
3. LMTD	1345°F
4. Heat load	1.085 x 10 <sup>9</sup> Btu/hr

Table A 11.6.1 Continued

Item	Specifications
Design Data - Convection Section (argon cooled)	
<u>Physical</u>	
1. Tube id	1.8 in
2. Tube wall thickness	0.1 in
3. Tube fin height	1.0 in
4. Fin thickness	0.030 in
5. Fin width	0.156 in
6. Tube material	SS
7. Tubing required	273,000 ft
<u>Thermal</u>	
1. Heat transfer coefficient - external	6 Btu/hr-ft <sup>2</sup> -°F
2. Fin efficiency	65%
3. Heat transfer coefficient - internal	100 Btu/hr-ft <sup>2</sup> -°F
4. LMTD	524°F
5. Heat load	2.41 x 10 <sup>9</sup> Btu/hr
Design Data - Convection Section (water cooled)	
<u>Physical</u>	
1. Same as argon convection system	
2. Tubing required	152,000 ft
<u>Thermal</u>	
1. Same as argon section (flue gas side)	
2. Heat load	1.34 x 10 <sup>9</sup> Btu/hr

Table A 11.6.2 Summary of Primary Heat Exchanger Design for Parametric Point 8

Item	Specifications	
1	Combustion Products flow	$8.69 \times 10^6$ lb/hr
2	Total heat load	$8.5 \times 10^9$ Btu/hr
3	Liquid sodium flow	$6.307 \times 10^8$ lb /hr
4	Sodium temperature rise	1481.3 - 1500°F ( $\Delta T = 18.7^\circ\text{F}$ )
5	Argon flow	$4.86 \times 10^7$ lb/hr
6	Argon pressure	1200 psi
7	Argon temperature rise	660 - 1500°F ( $\Delta T = 840^\circ\text{F}$ )

Design Data - Radiation Section

Physical

- |    |                     |  |
|----|---------------------|--|
| 1. | Tube id             | 2.0 in   |
| 2. | Tube wall thickness | 0.25 in  |
| 3. | Tube material       | Superalloy Haynes No. 188 -<br>with oxidation resistant cladding |
| 4. | Tubing required     | 76,500 ft  |

Thermal

- |    |                           |                               |
|----|---------------------------|-------------------------------|
| 1. | Heat transfer coefficient | 51 Btu/hr-ft <sup>2</sup> -°F |
| 2. | LMTD                      | 1397°F                        |
| 3. | Heat load                 | $5.0 \times 10^7$ Btu/hr      |

Design Data - Radiation-Convection Section

Physical

- |    |                     |                      |
|----|---------------------|----------------------|
| 1. | Tube id             | 1.5 in               |
| 2. | Tube wall thickness | 0.25 in              |
| 3. | Tube material       | Haynes Alloy No. 188 |
| 4. | Tubing Required     | 652,000 ft           |

Table A 11.6.2 Continued

Item	Specifications
Design Data - Radiation-Convection Section - Continued	
<u>Thermal</u>	
1. Heat transfer coefficient - external	12 Btu/hr-ft <sup>2</sup> -°F
2. Heat transfer coefficient - internal	112 Btu/hr-ft <sup>2</sup> -°F
3. LMTD	809°F
4. Heat load	2.9 x 10 <sup>9</sup> Btu/hr
Design Data - Convection Section	
<u>Physical</u>	
1. Finned tubes (same as base case)	
2. Tubing required	641,000 ft
<u>Thermal</u>	
1. Heat transfer coefficient - external	6.0 Btu/hr-ft-°F
2. Fin efficiency	65%
3. LMTD	190°F
4. Heat load	2.05 x 10 <sup>9</sup> Btu/hr

$$A_c = \left[ \frac{f \dot{m}^2 RT}{2g_c \frac{\Delta p}{p} p^2} A_s \right]^{1/3} \quad (\text{A 11.6.3})$$

where

- $f$  = fanning fraction factor
- $\dot{m}$  = mass flow rate of the gas
- $R$  = gas constant
- $T$  = temperature
- $\Delta p$  = pressure difference
- $p$  = pressure
- $A_s$  = heat transfer surface area

The characteristic length of the primary heat exchanger was determined from the known geometry of the tube (specified) and the previously calculated heat transfer surface and cross-sectional flow areas.

The cost the primary heat exchanger was calculated on the basis of the quantity of tube material, assuming the average tube weight was taken at 43.40 kg/m (6 lb/ft). This value is approximately 30% higher than the calculated weight of a finned tube of the geometry described in Tables A 11.6.1 and A 11.6.2, and approximately equal to the weight of the thick-walled tubing used in the radiation and radiation-convection sections of the primary heat exchanger design. The finned-tube portion of the heat exchanger represents about 70% of the total tube required. The weight factor used, therefore, was actually about 20% higher and was assumed to account for various struts and supports. The cost coefficient,  $C_{13}$  was taken as \$20.06/kg (\$9.10/lb) for the low-temperature case [922°K (1200°F)] and was derived on a weight averaged basis, from the following criteria:

- Radiation section - 1/10 of total weight at \$30.86/kg (\$14/lb)  
(Haynes clad SS)
- Radiation convection - 1/5 of total weight at \$30.86/kg (\$14/lb)  
(Haynes clad SS)
- Convection section - 7/10 of total weight at \$12.85/kg (\$5.83/lb)  
(316 SS)
- Tube fabrication cost - \$10/tube (10% of material cost)

The cost coefficients are used in Table 11.5.

For the high-temperature case, 1089°K (1500°F), the coefficient,  $C_{13}$ , was \$26.68/kg (\$12.10/lb), based on higher temperature materials in the final portion of the convection section. Approximately half of this section was assumed to require Haynes 25 tubing.

#### A 11.6.3 MHD-Steam Plant Coupling Heat Exchanger

Within the MHD system configuration, a heat exchanger is interposed between the MHD cycle and the steam bottoming plant wherein the waste heat of the MHD inert gas (argon) prior to compression is utilized to drive the steam turbine bottoming plant. The following sets of equations were derived to model the heat transfer and physical characteristics of this coupling heat exchanger for each parametric point.

It is assumed that the argon gas passes through the staggered tube array in crossflow. The heat transfer correlation appropriate to this situation is:

$$h = 0.33 \frac{k}{d_o} \left[ \frac{(v_{\max}) (\rho) (d_o)}{\mu} \right]^{0.6} Pr^{0.33} \quad (A 11.6.4)$$

This can most conveniently be expressed in the following form:

$$h, \text{ Btu/hr-ft}^2\text{-}^\circ\text{F} = (\phi_1) (v_{\max}^{0.6}) (d_o^{-0.4}) \quad (A 11.6.5)$$

So that  $V_{\max}$  might be expressed in ft/s units and  $d_o$  in inches,  $\phi_1$  must have the following form:

$$\phi_1 = \frac{(0.33)(12^{0.4})(k)(\rho^{0.6})(Pr^{0.33})(3600^{0.6})}{\mu^{0.6}} \left( \frac{\text{in}^{0.4} \text{ s}^{0.6} \text{ Btu}}{\text{hr-ft}^{2.4} \text{ }^\circ\text{F}} \right) \quad (\text{A } 11.6.6)$$

Values of thermal conductivity ( $k$ , Btu/hr-ft- $^\circ\text{F}$ ), density ( $\rho$ , lb/ft $^3$ ) and viscosity ( $\mu$ , lb/ft-hr) are evaluated at the arithmetic mean gas temperature.

The pressure drop per row for flow over a staggered array of tubes is represented by the following equation:

$$\frac{\Delta p}{N_{\text{row}}} = \frac{4f \rho V_{\max}^2}{2g \cdot 144} \quad (\text{A } 11.6.7)$$

where

$$f = \frac{0.507}{\left( \frac{V_{\max} \rho d_o}{\mu} \right)^{0.15}} \quad (\text{A } 11.6.8)$$

Equation A 11.6.7 can, therefore, be conveniently expressed as follows:

$$\frac{\Delta p}{N_{\text{row}}} = \phi_2 (V_{\max}^{1.85} d_o^{-0.15}) \quad \text{in psi/row} \quad (\text{A } 11.6.9)$$

and so that  $V_{\max}$  might be expressed in ft/s and  $d_o$  in inches

$$\phi_2 = \frac{(2)(0.507)(12^{0.15})}{(144)(3600^{0.15})(32.2)} \rho^{0.85} \mu^{0.15} \left( \frac{\text{lb s}^{1.85}}{\text{in}^{1.85} \text{ Pr}^{1.85}} \right) \quad (\text{A } 11.6.10)$$

For the liquid-metal MHD cases,  $\phi_2$  takes on the value 0.507, and the units are as shown in Equation A 11.6.10.

The required surface area for heat transfer per steam generator is given by Equation A 11.6.11:

$$A_s, \text{ ft}^2 = \frac{Q}{(N_{SG})(MTD)(h)} \quad (\text{A } 11.6.11)$$

Substituting Equation A 11.6.5 for h in Equation A 11.6.11, we obtain Equation A 11.6.12:

$$A_s, \text{ ft}^2 = \frac{(Q)(v_{\max}^{-0.6})(d_o^{0.4})}{(N_{SG})(MTD)\phi_1} \quad (\text{A } 11.6.12)$$

Equations A 11.6.13 and A 11.6.16, which follow, are fundamentally different ways of expressing the minimum area of flow per steam generator:

$$A_{f_{\min}} = \eta \frac{\pi D^2}{4} \left[ \frac{P_t - d_o}{P_t} \right] \quad (\text{A } 11.6.13)$$

In Equation A 11.6.13,  $\eta$  is a factor which expresses the fraction of the available area within the shell diameter (D) which is utilized for the tube banks. The tube pitch (P) is in inches, as is  $d_o$ .

For a situation where the tube banks are housed in a refractory metal duct of square cross section, which itself is housed within a cylindrical pressure containment shell,  $\eta$  has a maximum value of  $2/\pi$ .

Consequently, Equation A 11.6.13 can be restated as follows:

$$A_{f_{\min}}, \text{ ft}^2 = \phi_3 D^2 \left[ \frac{P_t - d_o}{P_t} \right] \quad (\text{A } 11.6.14)$$

where

$$\phi_3 = \left( \frac{\pi}{4} \right) \left( \frac{2}{\pi} \right) = 0.5 \quad (\text{A } 11.6.15)$$

The alternative way to express the minimum flow area is in terms of the mass flow rate of the gas,  $\dot{m}_g$ , and maximum velocity.



$$A_{f_{\min}}, \text{ ft}^2 \equiv \frac{\dot{m}_g}{(N_{SG})(V_{\max})(\rho)(3600)} \quad (\text{A } 11.6.16)$$

In Equation A 11.6.16,  $\dot{m}_g$  is the total MHD duct gas mass flow rate in lb/hr; hence, the need for  $N_{SG}$  in the denominator to put the flow on a per moduler basis. The 3600 in the denominator is required because  $V_{\max}$  is in units of ft/s.

Together, Equations A 11.6.14 and A 11.6.16 provide a means of expressing the inside shell diameter in terms of fundamental flow parameters.

$$\phi_3 D^2 \frac{P_t - d_o}{P_t} = \frac{\dot{m}_g}{(N_{SG})(V_{\max})(\rho)(3600)} \quad (\text{A } 11.6.17)$$

Letting

$$\phi_4 = (\phi_3)(3600 \rho) \left( \frac{\text{lb}}{\text{ft}^3 \text{ hr}} \right) \quad (\text{A } 11.6.18)$$

Equation A 11.6.17 can be conveniently rearranged as follows:

$$D = \left( \frac{\dot{m}_g}{\phi_4} \left[ \frac{P_t}{P_t - d_o} \right] \right)^{0.5} \left( N_{SG} V_{\max} \right)^{-0.5} \quad (\text{A } 11.6.19)$$

There exists yet another expression for the flow area, which is fundamentally different from those presented heretofore and which when equated with Equation A 11.6.16 will yield an equation for the length of tube per tube row (L). This equation is:

$$A_f, \text{ ft}^2 = \frac{L(P_t - d_o)}{12} \quad (\text{A } 11.6.20)$$

Equation A 11.6.20 is equivalent to saying that for every foot of tube length, L, there is a gap between tubes through which gas may flow, which is of width  $(P_t - d_o)/12$  ft.

Equating A 11.6.20 with A 11.6.16 we obtain:

$$L = \left( \frac{\dot{m}_g}{(P - d_o) \phi_5 N_{SG} v_{max}^{0.5}} \right) \quad (A 11.6.21)$$

where

$$\phi_5 = \frac{(\rho)(3600)}{12} \left( \frac{lb \ s}{hr-ft^2-in} \right) \quad (A 11.6.22)$$

In order to formulate an equation for the number of tube rows required we evolve a second equation for the heat transfer surface area.

$$A_s = (\pi) \left( \frac{d_o}{12} \right) (L) (N_{row}) \quad (A 11.6.23)$$

This is equated with a previous equation for  $A_s$ , namely; Equation A 11.6.12. Hence:

$$\frac{(Q) (v_{max}^{-0.6}) (d_o^{0.4})}{(N_{SG}) (MTD) (\phi_1)} = (\pi) \left( \frac{d_o}{12} \right) (L) (N_{row}) \quad (A 11.6.24)$$

Now substituting Equation A 11.6.21 for L in Equation A 11.6.24 and expressing for  $N_{row}$  we obtain:

$$N_{row} = \frac{12 (\phi_5)}{\pi \phi_1} \frac{Q (P_t - d_o)}{(MTD) (\dot{m}_g)} (v_{max}^{0.4}) (d_o^{-0.6}) \quad (A 11.6.25)$$

$$N_{row} = \phi_6 \frac{(Q) (P/d_o - 1)}{(MTD) (\dot{m}_g)} (v_{max}^{0.4}) (d_o^{0.4}) \quad (A 11.6.26)$$

In Equation A 11.6.26:

$$\phi_6 = \frac{12}{\pi} \frac{\phi_5}{\phi_1} \left( \frac{lb \ s^{0.4} \ ^\circ F}{in^{0.4} \ ft^{0.4} \ Btu} \right) \quad (A 11.6.27)$$

From Equations A 11.6.27 and A 11.6.9 we can obtain an equation for the overall pressure drop.

$$\Delta p = \phi_7 \left[ \frac{Q (P_t/d_o - 1)}{(MTD) (\dot{m}_g)} \right] (v_{\max}^{2.25}) (d_o^{0.25}) \quad (A 11.6.28)$$

In Equation A 11.6.28:

$$\phi_7 = \phi_2 = \phi_6 \left[ \frac{\text{lb}^2 \text{s}^{2.25} \text{ } ^\circ\text{F}}{\text{Btu in}^{2.25} \text{ ft}^{2.25}} \right] \quad (A 11.6.29)$$

When the tubes are in an equilateral staggered array, the height of the tube bank is given by:

$$H = N_{\text{rows}} \frac{P_t}{12} \cos 30^\circ = N_{\text{row}} \left( \frac{0.867}{12} \right) P_t \quad (A 11.6.30)$$

Substituting for  $N_{\text{row}}$  by Equation A 11.6.26 we obtain:

$$H = \frac{0.867}{12} \phi_6 P_t \frac{(P_t/d_o - 1) Q}{(MTD) (\dot{m}_g)} v_{\max}^{0.4} d_o^{0.4} \quad (A 11.6.31)$$

$$H = \phi_8 \left[ \frac{P_t}{d_o} \right] \frac{(P/d_o - 1)(Q)}{(MTD) (\dot{m}_g)} (v_{\max}^{0.4}) (d_o^{0.4}) \quad (A 11.6.32)$$

This value of H does not include any allowance for spacing between the various tube bank sections nor does it account for the height of the vessel heads. For sizing purposes, vessel heads were assumed to be hemispherical.

The analysis assumes the steam and water side heat transfer coefficients are large compared to those of the argon gas side, thereby neglecting steam-side thermal resistance. Furthermore, in the calculation scheme, the log mean temperature difference, MTD, was approximated by the

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arithmetic mean. Solution of the equations was obtained by specifying tube geometry [2.54 cm (1.0 in) od; 0.508 cm (2.0 in) pitch for all cases] and designing to a fixed pressure loss,  $\Delta p/p = 0.05$ .

The equations for calculating the heat exchanger weight are given in Table 11.5. The weight of the pressurized vessel,  $W_1$ , was calculated after using the hoop stress relationship for thin-walled vessels to determine the required shell wall thickness.  $W_2$  represents the total tube weight and can be calculated once the tube geometry and required heat transfer surface area are known. The costing of the steam generator was based on the calculated weight of both tube and shell, as indicated by the equations in Table 11.5.  $C_{14}$  and  $C_{15}$  are the appropriate cost coefficients for shell and tube, respectively.  $C_{14}$  was taken at \$1.46/kg (\$0.67/lb) and includes a factor for insulation.  $C_{15}$  depends on tube material and specific case and includes fabrication, which was taken at \$10/tube. The tube material costs, material type, and application to particular study points are summarized in Table 11.7.

#### A 11.6.4 Air Gas Preheaters and Recuperators

In cases where recuperator heat exchanger systems were utilized, the model for sizing was based on heat exchanger effectiveness, a concept widely applied in recuperative heat exchanger design. The equation expressing the required heat transfer surface area is:

$$A_s, \text{ ft}^2 = 815 \left[ \left( \frac{\epsilon}{1-\epsilon} \right)^{3/2} \frac{\dot{m} \text{ Pr}}{f_p} \right] \frac{RT}{g_c \left( \frac{\Delta p}{p} \right)} \quad (\text{A 11.6.33})$$

where  $\epsilon$  = effectiveness  
 $\dot{m}$  = mass flow rate, lb/s  
 $\text{Pr}$  = gas Prandtl number  
 $f$  = function factor  
 $p$  = absolute pressure, psf  
 $r$  = the gas constant,  $\frac{\text{ft-lbf}}{\text{lbm-}^\circ\text{R}}$   
 $T$  = average temperature,  $^\circ\text{R}$ .

This equation was derived from the heat and mass balance around the heat exchanger, the usual pressure drop expression; and used Reynolds analogy,  $h = f/2 \rho C_p V$ , to estimate the heat transfer coefficient. The equation is solved by specifying the effectiveness and friction factor.

For the air preheaters (argon precoolers), a similar approach to calculating the required heat transfer surface area was employed. For purposes of sizing and costing, the air preheater was taken to include both the air-to-argon exchanger and the air-to-flue gas stack-gas cooler. Equation A 11.6.14 was modified, therefore, to include a second term that accounted for the stack-gas cooler. This second term is given as

$$A_{s1} = 141 (MR) m \quad (A 11.6.34)$$

where MR is the air-to-gas ratio determined from an overall system heat balance. Equation A 11.6.34 is actually a simplified form of Equation A 11.6.14 where  $\epsilon$ ,  $p$ ,  $f$ ,  $t$  and  $\Delta p$  are fixed for all cases.

The expressions for calculating the characteristic size, weight, and cost for both the recuperators and air-gas preheaters were derived in a manner similar to that previously described in Section A 11.6.2. As before, tube geometry is fixed. For simplicity, 2.54 cm (1 in) od tubing was assumed. The derived size and cost algorithms are summarized in Table 11. . The cost algorithm shows two terms. The first term,  $C_9$  accounts for tube material costs, and the second term,  $C_{10}$ , accounts for fabrication costs taken at \$10/tube.

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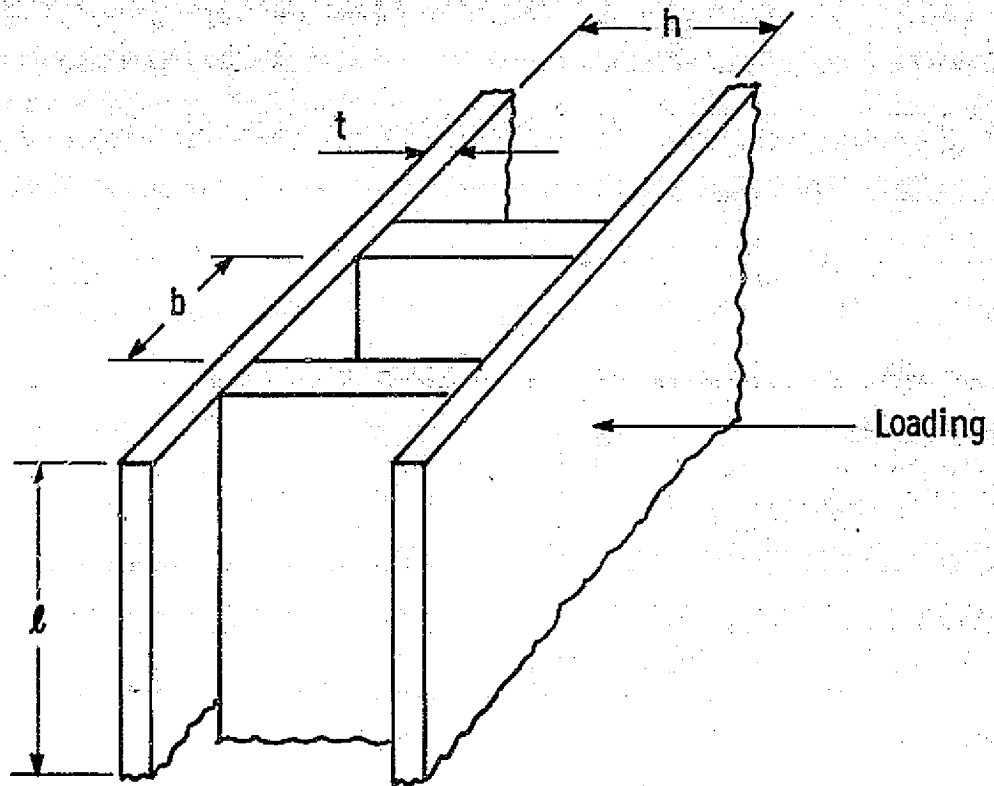


Fig. A 11. 7. 1 — Ribbed support structure for LM-MHD high pressure component design

## Appendix A 11.7

### STRUCTURAL REQUIREMENTS - MHD COMPONENTS

High system pressure, high temperatures, long operating life, and the large MHD component sizes result in structural requirements that must be properly addressed to assure a viable, economical, and consistent design. To meet these stringent design requirements, a basic ribbed support structure has been assumed, Figure A 11.7.1. The design approach for the pressure structure was to calculate a section modulus to accommodate the specific operating pressure and design stress conditions. The section modulus for the rib geometry used is approximated by (neglecting some second-order terms):

$$I/c \approx 4 bht/3 \quad \text{A 11.7.1}$$

where  $b$  is the spacing between ribs,  $t$  is the thickness of the plates and  $h$  is the spacing between plates (length of the rib).

For a beam with uniform load,  $p$ , on simple supports:

$$I/c = pb\ell^2/8 \sigma_d \quad \text{A 11.7.2}$$

$\sigma_d$  is design stress and  $\ell$  the unsupported length of the section. Assuming the web thickness is  $0.6 t$  and  $h$  is  $2 b$  (based on standard I-beam designs), the weight of the section per unit pressure surface area is given by Equation A 11.7.3:

$$W/A = 0.30 \rho p\ell^2/\sigma_d \quad \text{A 11.7.3}$$

where  $\rho$  is the material density.

This expression can be written as:

$$W/A = K \left( \frac{\bar{l}^2}{\sigma_d} \right)$$

where  $\bar{l}$  is in feet and  $W/A$  is in  $\text{lb}/\text{ft}^2$ , where  $K$  is a function of material and section geometry, and the remaining variables are specified based on the component size and operating conditions ( $p, \sigma_d$ ). With material density specified,  $K$  will depend on the rib dimension,  $h$ , which in turn is limited by practical size considerations (bending is a factor but is not significant over the range of sizes considered realistic for the MHD design purposes). For example, for the design of large MHD ducts [say 9.14 m (30 ft) long, 1.829 m (6 ft) wide], reasonable dimensions for  $h$  would be between 0.609 to 0.914 m (2 to 3 ft). For smaller duct (or other components) sizes,  $h$  would decrease. On this basis and over the range of component size for LM-MHD, the following expression for  $K$  was developed (assuming steel construction):

$$K = 155 - 17.5 \bar{l}$$

where  $\bar{l}$  is the characteristic unsupported dimension of the component in feet and  $K$  is in  $\text{lb}/\text{ft}^4$ .

This ribbed-plate support configuration served as the structural design member for two of the MHD duct walls and all remaining liquid-metal ducting and components, including the dewar for the magnet. The two MHD duct walls adjacent to the iron were designed assuming that the iron pieces (see Figure A 11.7.1) interposed between the MHD ducts served as load-bearing members. These duct walls were designed from flat plates assuming support I-beams at designated spacing along the plate. One of the flanged ends of the I-beams rests against the



iron pieces. The thickness of the plate forming the side wall is sufficient to support the load between I-beams and is given by fixing rib spacing at 0.3048 m (12 in):

$$t, \text{in} = 8.48 (p/\sigma_d)^{1/2}$$

The spacing between the I-beams is determined from physical constraints imposed by the design of the dewar.

The weights calculated for each MHD component were based on the geometry of the support structure described above. Weight equations and required sizing models for each specific component have been summarized in Table 11.5.