

USAAVRADCOM-TR-80-F-15

A COMPUTERIZED BALANCING TECHNIQUE FOR SUPERCRITICAL HELICOPTER SHAFTING

G. J. Korkosz HUGHES HELICOPTERS Division of Summa Corporation Culver City, Calif. 90230

December 1980

ELECTE FEB 1 8 1981

LEVFIT

Final Report for Period September 1978 - February 1980

Approved for public release; distribution unlimited.

Prepared for



APPLIED TECHNOLOGY LABORATORY POSITION STATEMENT

This report provides the documentation for a program which demonstrated a method for balancing a full-size supercritical helicopter tail rotor drive shaft using a quasi-static data acquisition technique. The process has been shown to place the center of gravity at all stations within 0.001 inch of the center of rotation. The balancing technique is available for implementation in existing and future tail rotor drive shafts.

Mr. Albert E. Easterling of the Propulsion Technical Area, Aeronautical Technical Division and Mr. Fred Reed of the Directorate for Systems Engineering and Development, US Army Aviation and Research Development Command served as project engineers for this effort.

DISCLAIMERS

The findings in this report are not to be construed as an official Department of the Army position unless to designated by other authorized documents.

When Government drawings, specifications, or other data are used for any purpose other than in connection with a definitely related Government procurement operation, the United States Government thereby incurs no responsibility nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission, to manufacture, use, or sell any patented invention that may in any way be related thereto.

Trade names cited in this report do not constitute an official endorsement or approval of the use of such commercial hardware or software.

DISPOSITION INSTRUCTIONS

Destroy this report when no longer needed. Do not return it to the originator.

	0
REPORT DOCUMENTATION PAGE	READ INSTRUCTIONS BEFORE COMPLETING FORM
USAAVRADCOM TR-80-F-15	TACCESSION NO. 3. RECIPIENT'S CATALOG NUMBER
A TITLE /Brd Suberties	S JUFE OF NEPONT & PENIOD COVER
A COMPUTERIZED BALANCING TEC	HNIQUE / Final Keport
FOR SUPERCRITICAL HELICOPTER	SHAFTING Sept 178-Feb 1980
	LEEBEORNING CRO REPORT NUMBER
7. AUTHORAL	HH-80-223 T
G I Korkog	DAAKEL 28 C AOLIN
G. J./KUROSZ	DAAK51-78-C-00211
9. PERFORMING ORGANIZATION NAME AND ADDRESS	10. PROGRAM ELEMENT, PROJECT, TAS AREA & WORK UNIT NUMBERS
Hughes Helicopters	1787132 (DEMA) 001 E
Division of Summa Corporation	1787123 (PEMA) 001 E.
11. CONTROLLING OFFICE NAME AND ADDRESS	TIE REPORT DATE
Directorate for Systems Engineering and Develo	opment [// Dec ember 1 0 80]
US Army Aviation Research and Development	Command 19. "NUMBER OF PAGES
St. Louis, Missouri 63120	53
Annial Take land the store the store	1
and Technology Laboratory, US Army Res	Search Unclassified
Fort Eustis, Virginia 23604	152. DECLASSIFICATION/DOWNGRADING
/·	
18. SUPPLEMENTARY NOTES	
18. SUPPLEMENTARY NOTES 19. KEY WORDS (Continue on reverse elds if necessary and identi	lfy by block number)
18. SUPPLEMENTARY NOTES 19. KEY WORDS (Continue on reverse elds II necessary and identional to the second	lly by block number) Drive Shaft
 SUPPLEMENTARY NOTES KEY WORDS (Continue on reverse elds II necessary and Identi Ultrasonic Gaging One Pass Balancing One La Balancing 	lly by block number) Drive Shaft Rotor
 SUPPLEMENTARY NOTES KEY WORDS (Continue on reverse elde if necessary and identi Ultrasonic Gaging One Pass Balancing 0.001 Inch Residual Eccentricity 	Ify by block number) Drive Shaft Rotor Supercritical Speed
 SUPPLEMENTARY NOTES KEY WORDS (Continue on tervices elde II necessary and identive Ultrasonic Gaging One Pass Balancing 0.001 Inch Residual Eccentricity AGETRACT (Continue on reverse othe N measuremy and identities) 	Ify by block number) Drive Shaft Rotor Supercritical Speed fr by block number)
 18. SUPPLEMENTARY NOTES 19. KEY WORDS (Continue on reveice elds if necessary and identi Ultrasonic Gaging One Pass Balancing 0.001 Inch Residual Eccentricity 24. ABSTRACT (Continue on every oth H measury and identify A full-size supercritical helicopter ta using a quasi-static data acquisition sonic gaging equipment and a digital location of the balance correction. I demonstrated the ability to locate the within 0.001 inch. The balanced sha 	Ity by block number) Drive Shaft Rotor Supercritical Speed Tr by block number) ail rotor drive shaft has been balanced technique. The method employs ultra computer to determine the amount and in one 15-minute pass, the system has e center of gravity (CG) of the shaft of the shaft.
 18. SUPPLEMENTARY NOTES 19. KEY WORDS (Continue on reverse elds II necessary and identi Ultrasonic Gaging One Pass Balancing 0.001 Inch Residual Eccentricity 28. ABSTRACT (Continue on covers etc. N mercenty and identify A full-size supercritical helicopter ta using a quasi-static data acquisition sonic gaging equipment and a digital location of the balance correction. I demonstrated the ability to locate the within 0.001 inch. The balanced sha 	If by block number) Drive Shaft Rotor Supercritical Speed by block number) ail rotor drive shaft has been balanced technique. The method employs ultra computer to determine the amount and in one 15-minute pass, the system has e center of gravity (CG) of the shaft off has run at speeds up to 10, 500 rpm,
 18. SUPPLEMENTARY NOTES 19. KEY WORDS (Continue on reverse elds II necessary and identi- Ultrasonic Gaging One Pass Balancing 0.001 Inch Residual Eccentricity 28. ABSTRACT (Continue on reverse elds N mercecery and identify > A full-size supercritical helicopter ta- using a quasi-static data acquisition sonic gaging equipment and a digital location of the balance correction. I demonstrated the ability to locate the within 0.001 inch. The balanced sha DD FORM 1473 EDITION OF 1 NOV 68 IS OBSOLETE 	If by block number) Drive Shaft Rotor Supercritical Speed by block number) ail rotor drive shaft has been balanced technique. The method employs ultra computer to determine the amount and in one 15-minute pass, the system has e center of gravity (CG) of the shaft off has run at speeds up to 10, 500 rpm, Unclassified
 18. SUPPLEMENTARY NOTES 19. KEY WORDS (Continue on reverse elds II necessary and Identi Ultrasonic Gaging One Pass Balancing 0.001 Inch Residual Eccentricity 24. ABSTRACT (Continue on reverse oth N measuremy and Identify 'A full-size supercritical helicopter ta using a quasi-static data acquisition sonic gaging equipment and a digital location of the balance correction. I demonstrated the ability to locate the within 0.001 inch. The balanced sha DD , COMM 1473 EDITION OF 1 MOV 68 IS OBSOLETE 	It's by block number) Drive Shaft Rotor Supercritical Speed Tr by block number) ail rotor drive shaft has been balanced technique. The method employs ultra computer to determine the amount and in one 15-minute pass, the system has e center of gravity (CG) of the shaft of the shaft of the shaft of the shaft of the shaft of the shaft of the shaft of the s
 18. SUPPLEMENTARY NOTES 19. KEY WORDS (Continue on reverse elds if necessary and identi- Ultrasonic Gaging One Pass Balancing 0.001 Inch Residual Eccentricity 28. AMSTRACT (Continue on reverse elds if metereory and identi- values a quasi-static data acquisition sonic gaging equipment and a digital location of the balance correction. I demonstrated the ability to locate the within 0.001 inch. The balanced sha DD FORM 1473 EDITION OF THOM ST IS OBSOLETE 	If by block number) Drive Shaft Rotor Supercritical Speed by block number) ail rotor drive shaft has been balanced technique. The method employs ultra computer to determine the amount and in one 15-minute pass, the system has e center of gravity (CG) of the shaft of thas run at speeds up to 10, 500 rpm, Unclassified SECURITY CLASSIFICATION OF THIS PAGE (Then Dele Br
 18. SUPPLEMENTARY NOTES 19. KEY WORDS (Continue on reverse elds if necessary and identi Ultrasonic Gaging One Pass Balancing 0.001 Inch Residual Eccentricity 24. AMSTRACY (Continue on coverse of the measure and identify > A full-size supercritical helicopter to using a quasi-static data acquisition sonic gaging equipment and a digital location of the balance correction. I demonstrated the ability to locate the within 0.001 inch. The balanced sha DD FORM 1473 EDITION OF 1 NOV 65 IS OBSOLETE 	If by block number) Drive Shaft Rotor Supercritical Speed by by block number) ail rotor drive shaft has been balance technique. The method employs ultra computer to determine the amount and in one 15-minute pass, the system has e center of gravity (CG) of the shaft of thas run at speeds up to 10, 500 rpm Unclassified SECURTY CLASSIFICATION OF THIS FAGE (From Date E

1 F

-1-1

,

Unclassified SECURITY CLASSIFICATION OF THIS PAGE(When Date Entered)

including sustained operation at four critical speeds, with acceptable vibration at critical speeds. Operation at the design speed (between 2nd and 3rd critical speed) was well within standard high quality balance values.

> Unclassified SECURITY CLASSIFICATION OF THIS PAGE (When Date Entered)

l B

SUMMARY

This report presents the results of a balancing technique for flexible rotors. The technique was demonstrated using the YAH-64 Phase I Helicopter Tail Rotor Drive Shaft as the flexible rotor.

The technique measures the mass distribution of the rotor about its center of rotation at a series of stations along its length. It uses ultrasonic gaging equipment, a digital computer to process the 50,000 data points measured by the equipment, and printout to define the localized balance needed to bring the local center of gravity back to the center of rotation. The process has been shown to place the center of gravity at all stations within 0.001 inch of the center of rotation.



PREFACE

This report was prepared by Hughes Helicopters, Division of Summa Corporation, under Contract No. DAAK51-78-C-0021, funded by Applied Technology Laboratory, USAAVRADCOM, Fort Eustis, Virginia. It covers the work performed during the period September 1978 to February 1980 and is the final technical report summarizing the activity. The contract statement of work breaks down the activities into six tasks.

TASK 1

A search was made for manufacturers of ultrasonic gaging equipment. Three companies were contacted for literature. All responded, but only one (N.D.T. Instruments) was able to delive 'hardware. It was concluded that the ability to obtain the measuring requirements for this program using ultrasonics was "state of the art".

TASK II

Design and fabricate the equipment required to modify an existing Hughes Helicopters' balancing fixture (used during the Phase I development of the YAH-64 Helicopter), for compatibility with the automatic data acquisition system and interface with the computer. Create the computer programs required to perform the calibration, acquisition, and data functions analysis. The final product is a plot and tabulation of the necessary balance corrections.

TASK III

Select two shafts that were manufactured during the Phase I development of the YAH-64 Helicopter Program. The two shafts (P/N 7-113500003) were S/N-1U, used in the Phase I YAH-64 ground test vehicle, and S/N-5U, a spare for air vehicles 02 and 03.

TASK IV

Trial balance S/N-1U for checkout and evaluation.

Balance S/N-5U witness by the Contracting Officer's representative.

TASK V

Conduct an economic analysis of this method for balancing 1,000 shafts.

TASK VI

Present a Government/industry briefing at Hughes Helicopters' facility, Culver City, California.

TABLE OF CONTENTS

.

	rage
SUMMARY	3
PREFACE	4
LIST OF ILLUSTRATIONS	6
LIST OF TABLES	7
INTRODUCTION.	8
INSTRUMENTATION	9
Ultrasonic Gaging	9 10 10 10
Digital Computer	11
SYSTEM CALIBRATION/RESOLUTION	12
PROCEDURE AND RESULTS	13
ECONOMIC ANALYSIS	17
CONCLUSIONS	18
RECOMMENDATIONS	19
APPENDIX A - TAIL ROTOR DRIVE SHAFT POSITION ENCODER	37
APPENDIX B - SUPER CRITICAL DRIVE SHAFT BALANCING SOFTWARE DOCUMENTATION AND LISTINGS	40

LIST OF ILLUSTRATIONS

,

Figure		Page
1	Schematic of balancing system	. 21
2	Schematic of balancing process	. 22
3	Ultrasonic signal	. 23
4	Computer plot of system calibration	. 24
5	Computer plot of shaft mass distribution S/N 1U	25
6	Computer plot of shaft mass distribution S/N 5U	. 26
7	First critical speed	, 27
8	Second critical speed • • • • • • • • • • • • • • • • • •	28
9	Third critical speed	29
10	Fourth critical speed	30
11	Calibration tube · · · · · · · · · · · · · · · · · · ·	31
12	Balance machine setup • • • • • • • • • • • • • • • • • • •	32
13	Nova scopes and transducer with transporter	33
14	Balancing tape use and application	34
15	Encoding equipment with reset control for computer interface.	35
16	Computer terminal and print/plot equipment	36
A-1	Drive shaft position encoder	38

LIST OF TABLES

,

Table		Page
1	Vibration characteristics - S/N-5U	14
2	Digital balance printout - S/N-5U	15

INTRODUCTION

As designers become more weight and vibration conscious for new aircraft designs, the introduction of power-transmitting shafts that run at supercritical speed becomes increasingly attractive. Balancing such high-speed shafts so that they run smoothly has long been a problem, with the process being both difficult and costly. It traditionally requires many trial runs over the rpm spectrum to achieve good balance.

The technique described in this report was developed by Hughes Helicopters (HH) and has been granted U.S. Patent No. 4, 170, 896 dated October 16, 1979. The balance method requires no trial runs to develop baseline data. All required data is obtained during a "one shot" pass along the slowly rotating shaft, is automatically processed, and the operator is given a graphic display of how much balance is to be applied along the shaft. The system is shown schematically in Figure 1.

Figure 2 is a schematic representation of the balancing process. The ultrasonic transducer and sensors ride along a track that is parallel to the shaft's centerline of rotation and measure two quantities at each of 250 azimuthal locations around the shaft for each lengthwise station along the shaft. The "slave" sensor measures the distance from the transducer to the outside surface of the shaft while the "master" measures the thickness of the wall of the shaft. The computer integrates this information around the periphery of the shaft, calculates the local center of gravity, calculates the amount and location of balancing tape needed to bring the local center of gravity and center of rotation into coincidence, and displays the balance requirements in both analog and digital form. The operator then applies the adhesive-backed aluminum tape to the surface of the shaft.

This method is valuable not only as a shaft-balancing procedure but also as a low-cost receiving inspection tool to qualify the basic tube for the important parameters which make an acceptable supercritical drive shaft, i.e., straightness, wall thickness variations, and dents. By rejecting tubes that require large or impractical balance adjustments, all the labor of shaft manufacture is saved, thereby minimizing the rejection rates for a finished shaft.

INSTRUMENTATION

The instrumentation required to conduct this program can be categorized into five basic areas:

- Ultrasonic gaging
- Preamplification
- Analog-to-digital conversion
- Shaft azimuth encoding
- Digital computation and output

ULTRASONIC GAGING

The ultrasonic equipment was purchased from N.D.T. Instruments, Inc., Huntington Beach, California. It consists of two specially prepared <u>Nova Scope 2000</u> instruments and one ultrasonic transducer with a water coupling adapter and a mechanical manipulator. One of the instruments acts as a slave, while the other performs as the master.

The master unit provides the excitation pulse to the transducer, and both Nova Scopes read back the ultrasonic signal response. The slave unit is calibrated to provide data on the "water path" or distance from the transducer to the top surface of the specimen (drive shaft surface). The master is set up and calibrated to provide the wall thickness data.

Both pieces of data are extracted simultaneously from the same ultrasonic pulse. Figure 3 shows a typical ultrasonic pulse and the interpretation of the peaks. The slave signal circuitry employs a blocking gate to delay the timing circuit. This gate is set up by the operator to some short time before the first echo. The purpose is to avoid false readings due to environmental noise. The time between the main pulse and the first return echo is equated to twice the distance to the first surface being measured. The instrument is calibrated by making precise known changes in the distance to the first surface, and adjusting the gain and zeroing potentiometers on the front of the Nova Scope. The digital display on the front of the Nova Scope provides a convenient indication for calibration and can be used at any time to indicate the parameter being gaged.

q

The master unit circuitry is set up to block out the main pulle and measure the time from the first return echo to the second return echo. This time is equated to the wall thickness. This unit is calibrated by using several known thicknesses of aluminum and adjusting the gain and zero potentiometers until the digital display indicates the thickness of the sample being gaged. An electrical connection to an analog voltage that is proportional to the digital signal displayed on the front is provided at the rear of each Nova Scope 2000. This analog voltage is the input to the computer.

PREAMPLIFIERS

Standard DC laboratory instrumentation amplifiers serve as preamplifiers between the analog output of the Nova Scopes and the analog-to-digital (A/D) converters of the computer to produce a digital resolution at the computer compatible with the desired sensitivities; i.e., less than 0.001 inch for eccentricity measurement and 0.0001 inch for the wall thickness measurement.

DIGITAL COMPUTER A/D

A Hughes Helicopters' laboratory computer is used for this program. It is a Hewlett Packard 1000 computer with a HP 2313 Data Acquisition System for multiplexing, A/D, timing, and control. The multiplexer is capable of data rates in excess of 40,000 readings per second. The azimuth error due to multiplexing 250 readings per revolution on three channels is 0.00675 degree, which is considered nil.

SHAFT AZIMUTH ENCODING

The data recording for this program needs to be referred back to a physical location on the drive shaft. This requirement is satisfied by the fabrication of a special Hugnes Helicopters developed electronic circuit that encodes the shaft into 250 equally spaced segments around its azimuth. The use of 250 units is an arbitrary but convenient buffer size in the digital computer.

The encoding system consists of an aluminum disc approximately 8 inches in diameter and 0.065 inch thick. The disc is machined to provide 250 equally spaced radial cuts 0.020 inch wide and 0.25 inch deep from the outside diameter. One of the cuts (the master) is 0.50 inch deep. A pair of infrared optical encoders are used to start and stop a pulse circuit. As the master slot in the timing plate passes the optical encoder, an automatic circuit is enabled, providing the computer with a trigger at each of 250 equally spaced azimuth locations. The automatic trigger circuit is disabled after one complete revolution, and no additional readings can be taken until the reset switch is activated. Figure 15 shows the encoding disc and the operator control switch. At each trigger, the computer scans the required data channels, stores the information, and waits for the next trigger. Appendix A provides a complete explanation and schematic of this system.

DIGITAL COMPUTER

The HP 1000 digital computer is programmed to do the following tasks:

- a. Calibrate the incoming voltages and convert the readings into engineering units.
- b. Read the data during a measurement run, place it into an array in memory, and print the results on a line printer after each azimuth scan (1 per station for 100 stations).
- c. Graph the results of a complete run on a plotter and print the required balance corrections in tabular form on a line printer.

Appendix B provides listings and explanations of all programs used for this project. Programs are written in both BASIC and FORTRAN. BASIC is used when speed is not a factor, while FORTRAN is needed to take data and perform the computations quickly.

SYSTEM CALIBRATION/RESOLUTION

The initial run of this system was checked against known inputs for reliability, repeatability, and accuracy. The first check assured that the voltages read at the analog output of the Nova Scope were being recorded by the computer. This check verified that magnitude, polarity, and channel designation were correct.

The second check verified that all the program logic and math algorithms functioned as expected.

The third check measured the wall thickness and eccentricity of a calibration tube (Figure 11) made specifically for this program. This tube was measured with micrometers to determine, by a separate method, the same parameters that the ultrasonic system does.

A fourth check was to run the system on the first drive shaft specimen, S/N-1U, and keep the transducer at the same position for all 100 station points.

This test proved system resolution to be less than 0.001 inch and 2 degrees for locating the centroid. The results of this test are plotted in Figure 4. In an ideal system, this check would show all data to be exactly the same.

PROCEDURE AND RESULTS

Two Phase 1 AAH shafts $(4.5^{\circ\circ} O. D. by 17 \text{ feet long})$ were balanced to demonstrate the technique. The first, S/N-1U, was used during Phase I development of the YAH-64 in the ground test vehicle and was not flightworthy because of some slight damage near one end, caused by rubbing on a frame. The damage is slight, a photograph would not show it, and the effect on the balance process is considered nil.

All balance material on the shaft was removed. The shaft was cleaned of all dirt or flakes of paint or epoxy that might cause an ultrasonic artifact and the shaft was installed in the spin fixture for the ultrasonic data scan. The first scan took 15 minutes, and produced a plot and tabulated correction weight chart within two minutes. The scan time is documented by the data acquisition rate of 8 seconds per station.

A laboratory technician was instructed in how to read the plotted and tabulated data. This instruction required approximately 30 minutes.

The hand lay-up of the aluminum balance tape used as the correction medium required 3.5 hours. The correction process is accomplished in two phases:

- a. Lay out, using a suitable marker, the azimuth locations (station 1 to station 100). This process was responsible for over 1.5 hours of the balancing time.
- b. Apply the tape in layers and lengths at the required azimuth, starting with the widest tape and working up to the narrowest.

The initial balance was not good enough, and upon investigation it was discovered that all computer corrections were based on 0.005-inch-thick aluminum tape, while the tape being used was 0.003 inch thick. A subsequent run of the same data with a new constant for the 0.003-inch-thicktape provided new data. A new taping operation was done in a 1.5-hour per period.

The operation of the shaft through four critical speeds (0-175 Hz) was successful, with success being determined by dwelling at the critical speeds for several minutes with no damper smoking or evidence of excessive heating. Figure 12 illustrates the shaft and mounting features.

The results of S/N-5U were similar to S/N-1U except only one taping was required.

The adhesive quality of the aluminum tape used in this demonstration was great enough to guarantee adherence to 5 times the maximum operational rpm of the YAH-64 tail rotor drive shaft. To keep the adhesive safe from environmental factors, a moisture-resistant scalant must be painted over the tape and especially along its edges. The density of the cured epoxy sealant is 0.47 lb/cu. in., and has been found to have little or no effect on the balance when used in the quantities required here.

Table 1 presents results of measured lateral response at the aft end of the shaft bearing mount. Data was recorded as velocity and converted to displacement and acceleration by analysis. This table is presented to show, quantitatively, the dynamic response characteristics of the drive shaft as mounted in the spin fixture. Comparisons to response characteristics when used in another environment, such as an A/C, may show significant differences due to the dynamics of the new structure attached.

Table 2 and Figures 5 and 6 are computer generated charts and graphs of S/N 1U and 5U showing mass distribution measurements recorded, and the required corrections. Figures 7 and 8 are spectrum analysis plots of before and after balance corrections on S/N 5U. Figures 9 and 10 are spectrum analysis plots after correction runs on S/N 5U. These figures show the pertinent vibratory response characteristics measured during these tests and are the source of the data presented in Table 1. The effect of the balancing process is clearly shown by comparison of the magnitudes of the various plots.

Frequency (Hz)	Voltage* (rms)	Velocity (in/sec)	Displacement (Inches Peak-to-Peak)	Acceleration (g's)
13	0.042	0.074	0.0018	0.016
50	0.142	0.741	0.0047	0.603
81**	0.0035	0.006	0.000024	0.008
106	0.110	0.194	0.0006	0.334
175	0.110	0.194	0.0004	0,550
*Voltage to engineering units = 0.00176 in/sec rms/mv rms				

TABLE 1. VIBRATION	CHARACTERISTICS -	S/N-5U
--------------------	-------------------	--------

****Operational** speed of driveshaft in YAH-64

TABLE 2. DIGITAL BALANCE PRINTOUT - S/N 5U

,

and the second se

.

1

аны: НЕСКОНКТ 6. 1900 НИМЕ: 518 РМ - S/N 5U

Call: FERRUARY 6. 1980 TIME: 5:19 PM Stotion 1 trach 1.2 trach 1.4

医过程检查 一些不	1330 [10].:	- 5 : 19	PM	
510110m	1 (no.h. 1 2	t and h	1.4 millione	Hnale (Dearee)
1	1	ы	ú	100
	1		50 134	176
•	1	11		187
.1	1	44	64 10	193
т. с.	1	1.1 1.1	10 12	191
	1	έ.	ហ្	171
r.	¥1	1	1	191
•	1	61	ů	212
:	1	0	ĥ	198
() ()	lái -	1	ί.	179
10	ы	1	й	144
11	ů	1	1	100
	0 0	1	1	133
,	i i i i i i i i i i i i i i i i i i i	1	1	148
4 .A 1.1	0		1	149
1.1	N G			121
1.1	0	11	1	62
1 +	Ŵ.	ł	н	50.
17	10 10	1	6.1	56
13.	ហ៊	1	1	87
1.14	()	1	1	4
. U	U	L	41	314
• 1	4	1	1	C
	ii ii	1	1	
	•,	1	1	3.
	1			D.)
4.11 207	1	•		54
C 1	•1	i	1	1.1
, 16-	L1	1	ì	,* * ,
	4	1	6.1	1403
. 1. 1	6 i	1	1	F.F.
je v	1	11		►.2
243	1	11	I	5.0
111		11		
3.4		ū	1	
	1		1	60
1.1	1	1	f 1	
	1		1	F.i.
	•		1	ni ni
36	1	Й	1	£.14
37	1	:	Ī	
38	1	i		- L.
र्ष	;	41	ů.	
40	2	Ú.	11	69
1.1			11	
•••	2	1	1.	· · · ·
4.	I.	1		L É ADA
·4 3	2	10 10	10 64	30 01
44	÷.	13	11	81
45	.'	ស្	'1	87

TABLE 2. DIGITAL BALANCE PRINTOUT - S/N 5U (CONT)

,

Sand Lote	L LEW E.	1 .2 Ann in	1 4 010 16 1010	Hards (16 mar)
200 190 190 190 190 190	to to to to	64 127 64 11 69	1 61 1 1	95 80 87 77
51 53 54 55	for the fact the fac	1ĝ 10 11 11 1	រ ស ស រ រ	82 83 87 85 87
50 50 50 50 50 60	$C_{\rm eff} = C_{\rm eff} + C_{\rm$	1 11 13 11 1	0 1 0 1 0	94 99 103 98
の1 の2 の3 の4 の5 。	to to set to to	1 1 5 1 1	4) 1 19 1 1	77 79 84 86 95
44 60 64 64 64	n to con	1 1 1 4 0	1 13 1 13 1	97 93 80 80 80 80
24 25 24 24 24	t data w u	19 65 64 1 1	19 29 1 1 1	2018 2019 역 2 역 2 역 2
2 2 2 2 1		1 1 1 1 1 1 1 1 1 1 1 1	t 1 	100 48 93 93 108
, 		1 	11 11 11 11	100 103 104 116 118
्र 		ា ស្ វ រ	41 J 43 41 41	123 122 122 134 116
14 42 44 44 44 44 44 44 44 44 44 44 44 44		1	1 7 13 1	118 107 97 99 128
47 95 199 199		1 1 1	អ ម ម មំ មំ	117 199 111 110 110

ECONOMIC ANALY SIS

A cost effectiveness analysis was conducted to determine the impact of using this method of balancing the YAH-64 supercritical tail rotor drive shafts.

The cost to install the system, including computer and programming time, is estimated at \$51,500. The estimate is in 1980 dollars and is based on the following equipment.

Manufacturer	Model	Name	Cost
HP	9825A	Desk Top Computer	\$12,000
HP	3437A	System Voltmeter	2,000
НР	Typical	Printer Plotter	5,000
HP	HP-IB	Interface Card	500
HP	A/R	I/O Roms	1,000
НР	3439A	Scanner	2,000
NDT	2000	Nova Scope	18,000
		Software Development	10,000
			\$51,500

3 F

The cost of this equipment, amortized over 1,000 parts, is \$51.50 each. The cost could be reduced considerably if the operation were installed at a facility that could time-share a computer. The actual C.P.U. time required to do this job is very short; however, during the data scan at a given station, the computer must devote continuous time to data acquisition.

The minimum equipment required is the ultrasonic system, the software, a high speed multiplexer, and the A/D converter. It is estimated that this minimum system could be installed for approximately \$35,000.

The time to actually balance a shaft can be divided into two areas: data acquisition and balance correction. The data acquisition and printout has been demonstrated to take less than 15 minutes. The balance correction has been demonstrated to take approximately 3 hours. These are actual documented times from the work done on this program; however, improvements of at least 2:1 can be realized under production conditions.

For comparison, the time to balance the YAH-64 tail rotor drive shaft using current technology is documented in program records at 8 hours per shaft.

Based on 1000 shafts, the labor saved is at least 3.75 hours per shaft. Assuming the value of labor through G & A to be 30.00 per hour, this represents a total labor savings of 112.50 per shaft, or 112,500.00 for 1000 shafts.

ĸ

The YAH-64 uses two supercritical and two subcritical shafts to which this process could be applied. Since four shafts per A/C represent four times as much balancing labor, as mentioned above, the costs for balancing 1000 shipsets of Tail Rotor Drive Shafts could be reduced approximately \$450,000.00, using this method.

CONCLUSIONS

It can be concluded after reviewing the data generated during this program that:

- a. The method described and demonstrated herein does locate the center of gravity of the shaft on the center of rotation, and it constitutes a condition of balance which is not changed by rotational speed.
- b. The degree of balance required is dependent on the desired operational speed of the end use of the shaft.
- c. A shaft designed to operate supercritically can be balanced by making one quasi-static data run. There is no requirement for high speed spin runs.

RECOMMENDATIONS

It is recommended that a study be conducted to determine the practical limitations of shaft interchangeability. Although a shaft can be balanced to locate its center of gravity within 0.001 inch of the center rotation, in a balance machine, the balance is only as good as the location of the center of gravity in its final use. This should be done during the original design of the shaft application, so that balance operations can be done on a bench set up during manufacture. The eccentricity of all mating rotating parts in the end application can sum to a significant unbalance which can only be accounted for in an in-place balance.

This program has highlighted the value of this technique as a receiving inspection tool, as well as a balancing technique. It is also recommended that this system be considered as a method of determining the acceptability of any tubular materials which are used in applications where straightness and wall thickness uniformity are essential.

Further development is recommended to extend the automation process, first to a computer-driven printer that locates the centerline of the balance tape along the drive shaft, and in a second step to develop an automatic tape-laying mechanism.





,

21



.

Figure 2. Schematic of balancing process.

-...



,

Figure 3. Ultrasonic signal.

بالاصطلاح الفاهينا سيحد بنبر المسادة الادر دسامية سعادهم





. . .





F





4 B 1.44



,

Figure 7. First critical speed (13 Hz).



Figure 8. Second critical speed (50 Hz).



,

Figure 9. Third critical speed (106 Hz).

1

and the second secon

29

.

250.00 Ŧ NO DAMPER CONTACT AT 175 H2, NO BEFORE BALANCE RUN DUE TO EXCESSIVE VIBRATION AT OVER 100 H2. 0.0 00.00 1 T Ţ 1 00.01 1 REAL

Figure 10. Fourth critical speed (175 Hz).

,

30

Ŀ

•



Figure 11. Calibration tube.



Figure 12. Balance machine setup.

٩



í,

Figure 13. Nova scopes and transducer with transporter.



Figure 14. Balancing tape use and application.



Figure 15. Encoding equipment with reset control for computer interface.



Figure 16. Computer terminal and print plot equipment.

APPENDIX A TAIL ROTOR DRIVE SHAFT POSITION ENCODER

Circuit Operation

The following description of the position encoder is referenced to the wiring schematic in Figure A-1.

Manual Mode

Upon power-up, a manual reset must be issued to set all registers to a ready state. The first 1/rev timing pulse (high) produces a corresponding high pulse at pin 8 of 74LS13(B); simultaneously, position 0 of the 250/rev produces a high at pin 6 of 74LS13(A). These simultaneous pulses cause register 74279(A) to become set, providing a "high" to pin 8 of 7408(C), and register 74279(B) to be reset which locks out all further set signals to register 74279(A) until a system reset is initiated. Each successive 250/rev pulse is then "added" with the set of register 74279(A) and fed to the transmission line where it is used to start and stop the HP computer pacer (H=run, L=stop). The next simultaneous 250/rev and 1/rev pulse will then clock the 7474(A) into a reset state which in turn resets register 74279(A), thus inhibiting all further pacer pulses until a system reset is initiated.

Automatic Mode

Automatic operation is basically the same as manual operation except a handshake operation between the encoding circuit and the computer is instituted and the "data in error automatic shutdown" circuit is enabled. In this case, the pacer S/S signal is disconnected from the transmission line and is first used to trigger the 74121 one shot, which in turn resets counter 7490 and register 7474(B). Secondly, the pacer S/S signal is "added" with the negation of register 7474(B) to form a new pacer S/S signal which is fed to the transmission line. When the pacer is started, pacer pulses are received via the transmission where they are counted by the 7490 counter.

After five pace pulses are received, the output of the counter will go high which sets register 7474(B), inhibiting the pacer S/S signal and stopping pacer. The pacer then remains off until the next 250/rev pulse and the sequence is repeated. Initial start-up and final shutdown are the same in the automatic mode as in the manual mode.

2.) ALL PASSUE CIRCUIT COMPOUNDENT SMOULD BE MOULTED VERTICALLY. I) + X = A THON TTL NEX ZUVERTER Nores:

-

3) Avoio Paravau wire Runs.

4) Supper Veltace is +546. 5) (M) = Manual (A) = Automatic Mode.

c.) AL IC FINE NOT SHOW ARE NOT CANARTED.
 c.) CAREAL CHECHT LIVONT IS REQUIRED FOR PROPER OFFERIND. WHER BUY LANOTH FROM T AS SHORT AF POSSIGLE A COMPAURUT ANET BE PAGEO AS GLORE AF POSSIGLE TO THERE COUNCINDS FONTS.
 c) THIS CLEALLY OFFERIAS AT FREQUENCY WATO IM HQ.



Figure A-1. Drive shaft position encoder.

,

In the automatic mode, the data error shutdown circuit is also enabled. Should the 250/rev timing pulse fall low before five pace pulses are received, the shaft travel will be considered too excessive for the data to be valid and the 7402 logic circuits will place the encoder circuit in a shutdown state, thus terminating operation until a system reset is initiated. Two internal dip switches are provided to switch between the auto/manual mode.

Data Rate

The data rate is limited by the response characteristics of the timing sensors and the external measurements to be taken. The rest of the circuit is capable of operating in excess of 1 MHz.

APPENDIX B

SUPER CRITICAL DRIVE SHAFT BALANCING SOFTWARE SOFTWARE DOCUMENTATION AND LISTING

The Drive Shaft Software package consists of four interrelated programs. The first is the Calibration program called DCAL. This program allows recalculation of the calibration constants which are required for the proper running of the programs which follow it. To accomplish this the program requires two aluminum plates of known thickness and "Eccentricity" (actually what is needed is the difference in heights of the two plates relative to the fluid nozzle).

After recalculation of the calibration constants, the program called DSHAFT is run. This program reads the calibration values from the disk and lists them on the CRT terminal. It then calls the two programs which actually accomplish the task of determining the amount of aluminum tape required to balance the shaft. These two programs are called DS1 and DS2. The first, DS1, reads in data from the Nova Scopes and calculates the cross-sectional area required to balance each of the 100 stations along the length of the shaft. The program then returns control to DSHAFT, which calls DS2. DS2 then determines the number of layers of each width tape needed to balance each station. Once these values are determined the program both graphs and prints the calculated tape information. When this program completes it returns control to DSHAFT which executes an END statement and turns over control to basic.

DCAL

After calling the subroutine NORM to normalize the 2313 data acquisition system, the program prompts the operator to input the value of wall thickness for the first calibration plate and the relative eccentricity of that plate. One of the two plates may be taken to have zero eccentricity and used as a standard value. The second plate then has an eccentricity relative to the first.

The program then waits until the operator is sure that the proper calibration plate is in place under the fluid nozzle. When the operator types "G" then the program begins taking two sets of 100 voltage readings. One set contains the voltage output from the Nova Scope for Eccentricity while the other contains the values for Wall Thickness. The 100 readings of each set are then averaged to produce one voltage reading for Wall Thickness and one voltage reading for Eccentricity. The program then notifies the operator of any voltage errors detected and of the completion of the first reading sequence. The operator is then asked to enter a new set of calibration values. These consist of the Wall Thickness and Eccentricity for the second plate. The program then waits until the operator types "G" before taking the next set of readings. This allows time to replace the first calibration plate by the second. Three sets of 100 readings are then taken and averaged to produce three voltage readings. These are the two Nova Scope readings plus a temperature reading. This third reading, the temperature, is used to account for variations in the readings due to the change in speed with temperature of the reflected signal.

Using the two sets of readings and calibration inputs, the program then calculates the slope and Zero voltage offset for both Wall Thickness and Eccentricity. These slopes have units of Volts per Inch (Volts/Inch) and are calculated using the equation for a line: $Y = m_*X + b$, where "m" is the slope and "b" is the offset. These two parameters are computed using the following two equations which also include temperature offset:

Slope: m = (input plate2 - input plate1)/(volts plate2 - volts plate1)

offset: b = (input platel) - (m*(volts platel)) - (temp. volts)/100

These values are then printed on the external printer and the values saved on the disk in file DSDATA. The voltage and input data are also stored along with the calculated slopes and offsets.

Note: The sign of the slopes is unimportant, as the program DSHAFT sets the sign of the slope to correspond with the observed behavior of the output of the Nova Scopes.

DSHAFT

6 F

> This program begins by reading the calibration values calculated by the program DCAL from the disk file DSDATA and displaying the values to the screen. However, prior to displaying the data the program corrects for the sign of the slope values by setting the Wall Thickness slope to a positive value and the Eccentricity slope to a negative value. The program then waits for the operator to hit Return before calling the FORTRAN subroutine DS1. When DS1 is called it is passed the calibration data from DSHAFT and returns the calculated values for Radius of centroid, Area required to balance the shaft, Angular location of the Centroid, Radius opposite Centroid, and First moment of inertia of the centroid. These calculated values are then stored on disk file DSDATA by DSHAFT. Next, they are passed to the FORTRAN subroutine, DS2, which completes the calculations. When DS2 completes and returns control to DSHAFT the program cycle is complete and DSHAFT ends.

DSI

This is the key program in the sequence. It is responsible for computing the location and magnitude of the imbalance at each station. To do this the program requires two calibration slopes; the wall thickness offset value at zero volts and the Radius of the shaft at zero degrees azimuth. Using these values the program takes 250 readings at each station at even angular divisions around the shaft. The readings are triggered by a change in voltage on a control channel which acts as a pacer for the readings. When the voltage on the control channel goes from a value below 2 volts to a value in excess of 6 volts the program takes one reading on each of three channels. The first channel is the Eccentricity voltage, the second is the Wall Thickness voltage and the third is the Temperature voltage. The program then waits for the next low to high transition before taking an additional reading. This continues for 250 readings around the shaft. (The program acts as a positive edge trigger device for data acquisition.) If after 250 readings it is determined that additional triggers are occurring on the pacer channel then an error condition occurs and the program prepares to retake the last set of readings after the operator hits the return key.

After the 250 sets of three readings have been taken the program calculates the location of the centroid and the centroid magnitude as follows:

a. First, the voltage reading for Eccentricity is converted into a Radius value and the voltage for Wall Thickness is converted into inches. Both these values have a correction for temperature variations by adding the temperature voltage divided by 100 to the inches value. (Note this is an empirical correction.) The centroid is then calculated by summing the individual moments about the circle. The individual moments are determined to be the area of that slice times the radius to the center of that area. The area is determined as

Area = (Wall thickness)*(Radius)*(Delta Theta)

This is a rectangular approximation to the area. The radius used is the radius determined by the eccentricity minus half the wall thickness.

b. The centroid is determined in Cartesian coordinates using the sine and cosine to determine the Y and X components of the moment vectors. These values are summed around the circle to determine the Centroid of the entire circular slice of the shaft. These values are then converted into Polar coordinates using R=SQRT((X*X)+(Y*Y)) and Theta=Arctan(Y/X). c. These values are then used to determine the amount of area required to balance the shaft at that station. The tape must be added at 180 degrees from the centroid of the shaft. This angle is determined and the actual radius of that location is calculated. This value is later used to determine the moment of a piece of tape at that location. The Radius of the centroid times the total cross-sectional area determines the magnitude of the imbalance. An equivalent moment must be added in the opposite direction in order to balance the shaft. The area of tape times its moment arm is this balance. Therefore, the area of tape needed = (Radius of the Centroid)*(Total cross-sectional area)/Radius opposite the Centroid).

DS2

This FORTRAN program is passed the information calculated by DS1 and computes the number of tape layers needed to balance each station. It then graphs and prints this data along with a graph of the radius of the centroid and the angle opposite the centroid.

The determination of the number of layers of each tape is based on an iterative process for the first size tape and then additional layers of smaller tape as needed. No more than one layer of the smaller layers is needed at any station due to a binary approximation method because each smaller layer of tape is 1/2 the size of the preceding layer. Therefore, only the tape widths 1 inch, 1/2 inch and 1/4 inch are needed.

To determine the effect of each layer of tape, the radius of the shaft at the balance angle is used and updated as additional layers of tape are added.

PROGRAM DCAL

DCAL T=00004 IS ON CR19513 USING 00012 BLKS R=0016

0001 TOPEN THIS IS THE DRIVE SHAFT BALANCING CALIBERTION PROGRAM "DCHL" 20FILESO, DSDATA ពិពិពិ? 30DIM 2023, WE 1023, EE 1023, TE 1003 0003 0004 40PEINT 0005 50CALLDATER(1) 0006 GOPPINT មួយថ្ម ? 76CALLNORM(1) SOPPINT 6668 SUPRINT "WHAT IS THE WALL THICKNESS"; 66699 100INPUT WE11 0010 0011 110PPINT ព័ណ៌ 🖓 120PRINT "WHAT IS THE ECCENTRICITY"; 130INPUT E[1] 6613 0014 140PRINT 0015 ISOREM THIS GOSUB WAITS UNTIL YOU ARE READY TO TAKE DATA 160G0SUB 800 0016 170LET Z[1]=0 180LET Z[2]=0 0017 មិមិ12 190LET E3=0 0010 2000ALLAISQV(100+-343,EL3]+E1) 8820 210CALLAISOV(100,-344,WE31,E2) 0021 200REM THIS GOSUB CHECKS THE EFROR RETURN VHPIHBLES EN 0022 250GOSUB 850 240IF E1+E2=OTHEN 270 250PRINT "CHECK INPUTS AND RETAKE FIRST SET" 0623 0024 0025 0026 2606070 160 0027 STOPPINT. 0020 200PRINT " FIRST READING SET DONE" 2006F0R I=1T0 100 300LET ZC11=ZC11+EC2+F1 310LET ZC21=ZC21+WC2+F1 0029 ំអូ ()អ្ 693. 0002 DEGNERT I 330LET 211=2111-100 340LET 2121=2121/100 350PRINT "NEXT SET OF PEADINGS" 350PRINT "WHAT IS THE WALL THICKNESS"; 350PRINT "WHAT IS THE WALL THICKNESS"; 00:33 6634 6605 9000 370INPUT WE21 3937 SUOPRINT "WHAT IS THE ECCENTRICITY"; 66.63 390THPUT CE21 6.539 400PEM THIS GOSUB WALTS UNTIL YOU ARE READY TO THEE DATA 41060508 800 4200ALLAISOV(100,~343,EL31,E1) 9941 0042 4000ALLATSOV(100.~344.0831.82) 0043 0044 +400ALLAISOV(100+-045+TL11+E3) 450REN THIS GOSUE CHECK'S THE EPROF RETURN VORTIONES 46060SUE \$50 0045 6046 4701F E1+E2+E3=0THEN 500 400PPINT " CHECK INPUTS AND RETAKE SECOND SUF" 0047 0048 490G0T0 410 0049 SUOPRINT "DONE, SECOND SET COMPLETE." 0050 S10FOR 1=2T0 100 520LET EC3]=EC3]+EC1+2] 0051 0052 500LET WE 3 J=WE 3 J+WE I+2 J 540LET TE 1 J=TE 1 J+TE I J 0053 8054 SSONEXT I SEOLET E[3]=E[3]/100 6055 0056 9957 570LET NECO =WE 3 1/100 0058 580LET TE1 J=TE1 1 10000 SHOPEN T IS THE TEMP CONSTANT IN INCHES CHANGE 6059 0060 600LET W=(WC23-WC13) (WL33+CC23)

PROGRAM DCAL (CONT)

610LET E=(E[2]-E[1]) <(E[3]-2[1]) 00610062 620LET E0=E011-2011+E-T011 0063 630LET W0=W011-2021*W-T011 640PPINT #6 650PPINT #6:" THE FOLLOWING CONSTANTS WERE TAKEN ON:" 0064 0085 3666 3967 660CALLDATER(6) 670PRINT #6 EXPERINT #6; "WALL THICHNESS CONSTANT (Inches Wolt) = ":W ESOPRINT #6; "WALL THICHNESS OFFSET (Inches) = ":W0 700PRINT #6; "ECCENTRICITY CONSTANT (Inches) Volt) = ":E 710PRINT #6; "ECCENTRICITY OFFSET (Inches) = "E0 720PRINT #6; "TEMPERATURE OFFSET (Temp/1000) = ";T[1] 730PRINT "OK (Y or N)": 0068 00690067 0070 0071 00772 00773 00770 00770 00770 00770 00770 00770 00770 (Temp/1000) = ";T[1] 740INPUT H# 750IF A#="N"THEN 90 7509FUNT #2:1 7709FUNT #2:E0:N0;E:N:NC1]:NC2]:EE1]:EE2]:TE1]:ZE1]:ZE2]:EE3]:NE3] 780CHAIN "DSHAFT" 790FEM WAIT GOSUB 3079 6380 -800PPINT "When you are ready to take data type G and Retrun." SIGINPUT HI 3081 S201F AS#"G"THEN S10 S30RETURN 6983 . . . ê(i94 -840REM VOLTAGE ERROR SUBROUTINE SSOIF E1#0PPINT "Voltage error on Eccentricity reading Channel 343" 860IF E2#0PRINT "Voltage error on Wall thickness reading Ch. #344" 070IF E3#0PRINT "Voltage error on Temperature reading Channel 345" 880PRINT "VOLTAGE ERROR CHECKSUM = ";E1+E2+E3 0005 30.06 0387 0033 SPOPRINT 8085 0.000**200RETURN** 0091 910END

PROGRAM DSHAFT

DSHAFT T=00004 IS ON CR19513 USING 00012 BLKS R=0016 10REM THIS IS THE DRIVE SHAFT BALANCING PROGRAM "DSHAFT" 0001 0002 20FILES0, DSDATA SOREN WE + 3=BWGHTE + 3, HE + 3=ATHETAE + 3, RE + 3=PBAP+E ECC, B=BAL ANNA & 40DIM EC 100], 80 100], WC 100], RC 100], AC 100] មិមិមិមិ4 0005 50READ #2+1 មិចិចិត 60FOR I=1TO 70READ #2;ECI] 0007 SONE::T I <u>ព័ល៌ពិ</u>ន **90REM** 0009 100PEM THE FOLLOWING TWO ASSIGNMENTS HEE TO ACCOUNT FOR THE 0010 100EM FACT THAT THE SLOPE OF THE ECCENTRICTLY VOLTS SHOULD BE 120REM NEGATIVE WHILE THE SLOPE OF THE WALL THICKNESS VOLTAGE 00110612 ISUREM SHOULD HAVE A POSITIVE SLOPE 140LET E[S]=-ABS(E[3]) 6613 ññ14 150LET EC41=ABS(EC41) 160PPINT "OFSET VALUE FOR ECCENTRICITY= "REL11 3915 0016 TROPRINT "CCENTRICITY SLOPE'IN VIENTITY "(ELS) 180PRINT "OFSET VALUE FOR WALL THICKNESS= "(ELS) 190PRINT "WALL THICKNESS SLOPE (In V)= "(EL4) 0017 0018 6619 200PRINT " CALIBRATION VALUES USED" 0020 0021 210PRINT 220PRINT "WALL THICKNESSES USED : FIPST = ":EL5];" SECOND = ":EL6] 230PRINT "ECCENTRICITIES USED : FIRST = ":EL7];" SECOND = ":EL8] 0022 230PRINT "ECCENTRICITIES USED : 6623 240PRINT "VOLTAGES READ : FIRST SET SECOND SET" 0624 2025 250PRINT 6026 260PRINT "CHANNEL 343";TAB(18);EF103;TAB(35);EF123 270PRINT "CHANNEL 344"; TAB(18); EC 11]; TAB(35 (; EC 13) 0027 6638 280PRINT 8629 200PRINT "TEMPERATURE OFFSET";E[9] 0030 3031 300PP INT **310PRINT** COOPPINT "WHEN READY HIT RETURN" 0032 330INPUT AF 0033 6034 340CALLDATER: 1 · 0035 350CALLDS1 (EC11, BC11, WC11, RC11, AC11) 360CALLDATER (6) 0036 0037 3708EAD #2,2 380FOP I=1T0 100 0038 0901F 1=51READ #2+3 6039 400PRINT #24ECIJ 0040 0041 410NEXT I 420READ #2+4 0042 3843 430F0R I=1T0 100 0044 440IF 1=51READ #2+5 450PR1HT #2:8013 0045 0046 460NECT 1 0047 470READ #2+6 480FOR I=1TO 100 0048 4901F I=51READ #2+7 0049 500PRINT #2; W(1) 8650 SIGNEXT I 0051 520PEAD #2+8 0052 530FOR 1=1TO 100 0053 5401F 1=51READ #2.9 6054 SSOPPINT #2:REI1 0055 0056 560HENT 0057 570PEAD #2-10 580F0P 1=1T0 100 0058 5901F 1=51READ #2+11 6659 600PPINT #2;AE11 9960 0.061610HEDT I GROPHUSE 0062 0063 GB0(ALLDS2(EE1),BE1),WE1),PE1),AL1); 640END 0064

· •

PROGRAM DS1

DS1	T=6	00004 IS ON CR19513 USING 00038 DLKS P≕0294	
6661	СТЫ	4.1	
0001	P 119	496. - споролитис вся сах вон рисит рвов атиста.	
000		SUBRUUTINE DST EULOBELOWGHONDRONTHLING	
0003		DIMENSION VOLISIS, 250 (VV) 5) (EC) 100 (GR) 100 (GR) (TTCS) (MTMCS)	
0004		DIMENSION BUGHI (100) REHECION ATHETH 100)	
0005		DATA PI/3.141592654/	
0006		DATA INOPI/6.2831853/	
0007		DATA DTHETA 1.02513274	
0008		DATA ISTATZ1/	
8889	0	THIS PROGRAM CALCULATES THE CENTROID OF H PLANAR SLICE OF A DRIVE	
0010	C	SHAFT TAKEN NORMHL TO THE AXIS OF ROTATION.	
0011	C		
3012	C	THE DATA IS TAKEN IN POLAR COORDINATES AND THEN CONVERTED INTO	
0010	C	CARTESIAN FORM USING THE RELATIONS:	
0014	C	N=r€05/theta→	
0015	C	Y=rS1H(theto)	
001 H	Ç	VOLTS(1,1) WILL HOLD THE VALUES FOR FOC(191A) (CHTRICITY	
2017	0	VOLTS(2.1.1. WILL HOLD THE VALUES FOR WHILL THICKNESS	
2018	Ú.	VOLTS(3,I) WILL HOLD THE VALUES FOR WATER TENREPATURE	
0.1.	0	PADIUS WILL HOLD THE LOCATION OF THE MG FOR THE SECTOP	
2020	Ū,	AREA NILL HOLD THE CALCULATED AREA OF CACH SECTOR	
0.021	C		
- 1	C	THE FOLLOWING ASSUMPTIONS ARE MADE:	
0320	C	1. THE SECTOR IS APPOX. A RECTANGLE	
1.1	Ē.	2. THE CENTER OF MASS, MAR FOR THE SECTOR IS HALF	
	C .	THE WALL THICKNESS FROM THE OUTER EDGE.	
	1	3. THE SECTOP HAS UNIFORM DENSITY	
3027	C		
10120	Ę.	XA IS THE SUMATION VARIABLE FOR X MULTIPLIED BY AREA	
ER DE	C.	YA IS THE SUMMTION VARIABLE FOR Y MUUTIPLIED BY AREA	
3050	1	A IS THE SUMATION VARIABLE FOR THE OREA	
	C		
	Ú.	THIS BEGINS THE DUMMY ARRAY VALUE SECTION FOR DEBUG	
	C.	THIS BEGINS THE INITIALIZATION SECTION FOR DEFINING CONSTANTS	
s. 1 (14		WRITE(1+2)	
		READ (1.+) IPRINT	
		WRITE(1+3)	
		READ (1++) PCON	
		RCON#RCON+2	
- 200 P		WPITE 1,13	
់ភ្នំ4វ		READ (1,+) PF	
· · · + 1		V1 = ECU(1)	
1114		CON1=ECC(+3)	
- 1 - 1 - 1 		M2=ED(+2+	
0.144		LUNA2#EUU/44	
1145		WRITE 1.1. CONT. CON2. RCON	
- 95° A		PMH2=6.0	
		PM [1]=2.0	
11.14.1	Ľ_	THIS IS THE DATH HEQUISITION LOOP	
0045		CHEL NUMBER STATES	
ណ៍ទីល	Ē	CHLL EBECTITITITIC TABLE	
9921	1	DO 50 IEI-200	
JH	25	(
0050		IF (FYUE), GI.F.MIN (JULU 2)	
មូលភ្នុង	36	CALLENDER IN AND AND A CONTRACT AND A CO	
90		IF (FYUELLE), FNHELS (JULU LE)	
-111 (r.		ENEL NATURE EST SAUE EST SAUE EST AND	
	50	1 CONTRACTOR A CONTRACTOR A	
005-		UNLL FREEFELA ANALALEFE	
1.15		1 日1 七、日前1947年(十月26年)(中国地区)(七世世世)	

47

g esterio de e

. ...

PROGRAM DSI (CONT)

,

0060	CALL PACER(0.0.0.1.IEPP)
0061	VMIN=V(1)
3062	(M68=2)(1)
0060	DO 27 I=2,75
2064	IF (V(I).LT.VMIN(VMIN=V(I)
0065	IF (V(I).GT.VMHX) VMHX=V(I)
9066	27 CONTINUE
110	IF (VMMAX-VMIN) .LT. 2, GOTO 20
(Uo)	MR1/EV1+247 15/14/
	MEHD (1,*) HHL1
	U UHLL EXECUTIONITA TREMES
	A ANINE ANINE ANINE SETTING AN ANIMALIA ANIMALIA ANIMALIA ANIMALIA ANIMALIA ANIMALIA ANIMALIA ANIMALIA ANIMALIA ANIA ANIMALIA ANIMALIA ANIMALIA ANIMALIA
	U SEC=AMIN+M(10/2)-1/1(2) + ((MTIM(1)-1))(1)(100.0)
9974 0075	U METTE (5320/350U) C CT CUNCTION PORTOUTO TO PERO
0070	C SET SOMETION VERTIBLES TO ZEPO
0.075	25 AN-0 Molo
0075	
- K976 3676	
3029	U 7 THIS BECING THE COLONIATION SECTION
0000	Contra begins the checkention SECTION
0001	с V1=V0(ТС:1:1:=C0N1
0002	
00000	IF (PE FG 1) WEITERS, (2) I.000 FCCC. I.000 FCCC. F. 0.5 F. 0.5
5004	WINTS(1, $1 \ge 0.00 + VOLTS(1, 1) + 0.00 + 0$
900.0	
0087	Replus=Vol TS(1,1) - (Vol TS(2,1) -)
0033	AREA=VOLTS(2+1)+PADIUS+DIALIA
0085	IF (PF.E0.1) WPITE(6,19) YOUTS(1,1), MODELS(2,1), PRODUCT, APRO
0090	THETA=(I+1)+DTHETA
0091	28= 28+,88+8+885105+C03+T0576++
10.10	YA= YA~(AREA+RADIUS+SIN(THETA))
0093	A= A+APEA
មួលមិត	100 CONTINUE
0025	XBAR= XAV A
0096	YBAR= YA & A
0097	RBAR(ISTAT)=SORT():DAR+XBAR)+(YBAP+YBAR))
0098	THETA=ATAN2(YBAR)(BAR)
0099	C
0100	C BEGIN PRINTOUT SECTION
0101	THETA#THETA*360/TWOPI
0102	IF (THETA .LT. 0) THETA=360+THETA
0100	WPITE(IPRINT,7) ISTAT
0104	WRITE(IPPINT+9)
0105	WEITE IPPINT 10 (PRAP) ISTAT
9106	WFITE IFFINI . 11 THETA
010	WEITERIPHINI 12 CH
0100	TE CHELH LL. 1980 HINGLE (THE TRACES)
0102	THE CHELH LEE. 1000 HIGLEFIHE (M-1000.
0110	
011 011	THELE THINGLE CONTRACTOR STATE
0112	атор постата солжении Птерион стратовать а тыргатат.
0114	
0115	IANGLE=1THETA+1
0116	IF (IANGLE .G). 2500 1006(€≥)
0117	DELTA=VOLTS: 1. IANGLE - VOLTS: 1. INDONES
0110	ECC(ISTAT)=VOLTO(1+ITHETA+++)NET N++NET
0119	WRITE IPPINT 14 CTHETA LECCLISTAL
0120	ATHETA ISTATIC DETA
0121	BALKISTAT (#6+D) ARKINIA
0122	BNGHT+1STAT+=EAL+1STAT+ LECC+1STAT+

PROGRAM DS1 (CONT)

,

0123		WRITEVIARINT,21 · BWGHT(ISTAT)
0124		IF (ABS(RBAR(ISTAT)) .LT. 0.0001 (URITE) (PRINT,15)
0125		ISTAT=ISTAT+1
0126		WRITE(IPRINT,16)
0127 -		IF (ISTAT .LE. 100) GOTO 1
8128 -	2	FORMAT("WHAT IS THE PRINTER (1 OR 6)?")
0129	3	FORMAT(/"DIAMETER IN INCHES?"
0130	4	FORMAIX"Eccentricity Constant (Income Volt)"))
0131	5	FORMAT(, "Wall Thickness Const. / Inches Volt)?"./
0132	6	FORMAT(,"Woter Temperature ====================================
6133	7	FORMAT("STATION NUMBER "+I3)
0134	9	FORMAT("CENTROID IN POLAR COORDINATES:")
0135	10	FORMAT(" RADIUS IN INCHEC = "+FOL++>
0136	11	FORMAT(" DEGREES A2IMUTH = "((5.2)
3137	12	FORMAT(SX: TOTAL R-SECTIONAL AREA = (F10.4)
913 <u>8</u> -	10	FORMATY DO YOU WANT A DATA DUMP (1=YES)(")
0134	14	FORMATC ADD WEIGHT AT "(F5.2)" DEG ACIMUTH. RADIUS 2"(F0.4)
0140	15	FURMATIC ** BALANCED **")
0141	16	FURNHI ()
0142	17	FURMATICTEUC CONSTET, F8.4/ WHEL CONSTET, F8.4/ WHEL CONSTET, F8.4/
0143	18	FURMHICLEUE V. 474F8.54 WHEL V. 474F8.54 TEMP V. 774F0.14
0144	19	FURNHILLEUUETERS, S.T. PHILEESERS, S.T. PHILUETER, S. THUMETERS, S. T.
0140	<i>4</i> 1	FURMELY HEADSTOLENING MENDUCES FOURINITY'S
0140	22	PUPPUSITE CENTROID/= (F7.5) DEPMOTIVE CENTROID/= (F7.5)
0147	<u> </u>	FORMALL WHAT IS THE OFFSET VHEOR FOR ECCENTRICITY ()
0140	20	FURNITION WORL IS THE UPPORT (2010) FOR WHILE (MIUKNESSING) Formations (2010) - Magine Thank Sea and the Control (2010)
0142	24	FURNMENT CREWET, NURGE HUNG 200 (0.30) VELLEVE. /
0150		A RESEL FOR STATION (13) THEN ALT RETURN.) Dettem
0152		
0152		
	-	

7 F

PROGRAM DS2

,

7 B

DS2	T≍00	0004 IS ON CR19513 USING 00025 BUIS P+0.00.
0001	FTH4+	L
មិមិមិដ្ឋ		SUBROUTINE IG2/ECC+BAL+BAGH1+Runk+AftH1 to -
0003		DIMENSION ()+D++IL+S++IV+11++ECC+1400+4446+1400+
មិចិត៌4		DIMENSION ERGET-100+FEGEP100+FIHE16-100+FTH1FF1005
មិចិចិត		DIMENSION ANTAINDA
9996 -		INTEGER TOT(100+3)+BLIC+RTIC
ðbð?		DATA BMAC 0.0.
0005		DATA PMAC 0.0
មិ <u>មិមិ</u> ម		DATA ISTAT 1
9919 -		DATA INVIA-2007-2007-2010-2000-2000-2000-2000-2000
ê011		BATA IY/20,2HDH,CHUA,2HNU,2HE,CHUE,CHIG,CHHT,CH,H,CHPE,CHA)
3912 -		
6913		$DO_2 I=1,160$
0014		$\mathbf{ICI}(\mathbf{I},\mathbf{I}) = 0$
0015		ICT(1+2)=0
1116		101(1:3)=0
55 I T		AHIYYY AN ANY ANY ANY ANY ANY ANY ANY ANY A
1913		IF (REHE)I, GI, FOH () ROBLERDIEN I
		18 (BNGM)'1',G(,BMM,) BMM,⇒BNG,=CNGM()'1' CODETENS
1020		
- 양양교 I - 고려 전	1.2	EDEMOTING' FORMATING'S States I wash to seek the wash Theory (States)
100 a.u. 100 a.u. 100	1,	- NORMET 1077 - 2001 101 - 1 1001 - 1 2 1000 - 1 4 1000 1000 (2000) 2010 - General Demonstration
0.20	1.8	*
0023	10	BTIC=10
11.1		BSCALE=300
		IF (BNAC .LE01) GOTO 200
		BSCALE+150
$0 \in \mathbb{Z}^{n}$		IF (BMAN .LE02) GOTO 300
$1 \leq i \leq i \leq n$		BSCALE=60
0.031		IF (BNAX .LE05) GOTO 300
9657		ETIC=15
0.033		ESCALE+40
मन्द्र प्रम करुका		IF (BAAR LE0.5) 6010 300
100		
		550025-50 Teleday e il coto 200
non o Grafier		AF (DOMA (DE,)I' GOTO ODO RTICE:5
111		5 (1 - 1 - 2 - 1 - 2 - 1 - 1 - 2 - 1 - 2 - 1 - 2 - 1 - 2 - 2
0040		IF (BMA) .1715. 5010 300
0.14		ETIC=10
304.1		BSCALE=15
<u>0043</u>	300	PTIC=10
5044		RSCALE = 150
-9945 1		IF (PMAC).LE02 (GOTO 500
0046		RSCALE = 60
0047		IF (ROAN LE. 1957 GOTO 500
0048		R/11(=1) Received (= 10
0049		ROUNDEANU Te administration activity and
0000		tr (Kana), je, toto Gold and Diffelia
- 1997) 1. 665 1		en alemane Recipie e com
005		$\mathbf{F} = \mathbf{F} = \mathbf{F} = \mathbf{F} = \mathbf{F} + \mathbf{F} + \mathbf{F} = \mathbf{F} + \mathbf{F} = \mathbf{F} + \mathbf{F} + \mathbf{F} + \mathbf{F} + \mathbf{F} = \mathbf{F} + $
6654		RTI(=15
010 F 1		RSCALEFOO
1111-1		IF (PD6.) .LE15 - 6010 500
1.11		RTIC=10
it i ^e		PSCALE=15
- in 1974		IF (RNAC),LE2 (GUTO NOO
19-11		P 30 ALERV

PROGRAM DS2 (CONT)

	20200	CHEL PETERDIA
		CALL LEAFT
C 252		CALL SEACT-15. (10.)
d training in		C着LE ₽107(A.↓A.∀ №)
1 - C		CHEL PLOTE S. S. S.
		THE TOME STUDY
		10 BIN 12
		사망 교육은 유민을 만드셨다.
•		
		9945, STRES, STIN, 1, 13, (90, (-))
· · · •		CONTRACTOR *
• 1		CALL SYME(.1+3,5+,1+11,+0,+ 2)
		100 520 141・3
		21年1日+(114+1)
N 74		CAEL SYME(11:3.5.1.1.1
1.1	520	CONTINUE
2174		Church (1914) - Church (1955) - Church (1955) - Church (1955)
* 4 ×		LINE STUDIES FOR
		- 「「「」」」」」」」」」」」」」」」」」」」」」」」」」」
		ALTERNIS CONTRACTOR DULLS IN A RECENT AND A RE
1. 	£	- 5.0120 2000000000000000000000000000000000
	9.56	STATES AND A STATES
N 1775 N		CHEL SYME(31+.5+.1+13.+0.+.2)
$\{1,1,2,\dots,n\}$	535	CONTINUE
qua Sec		CTIC=1
-111 ·		SI=.64
da Con		CALL SYMBER, 6, 644, 54, 1415, 46, 4414
(1, 2)		90 SBE 1-4411
1.54		CALL HUMER CITY, SALIATION AND A REP
÷ 1.04 .		UTIC=I+40
11		11 I = 1 + 1 - 3 + - 775
	c , 1 ₂ ,	СБИТТЬБЬ СБИТТЬБЬ
		TABLE COMPLETE AN INTERATION OF
1111-1		THE STREET STREET AND AND AND AND A STREET AND A ST
naas		- SHEE SHIDESSERVESSERESSERESEESEESEESEESEESEESEESEESEESE
0070 0007		10 300 1-000
0024		11年122日本に14日1日。
0090		(LIU#I+W.E-BELHLE
0099	F • •	(AEL_NUMBER, 4+11, 1+1110,90, ,+3)
0100	538	LONTINUE
0101		CALL PLOT(.5,3.5,3)
0102		DO 540 I≈0×PTIC
0103		VI=3.5++1+3. PTIC+
0104		CALL SYMB(.5.YI1,1090.,-2)
0105	540	CONTINUE
9106		CALL SYME: 1.6.5.1.1.11
0107		DÚ 550 I≃0.10
0130		XI=.5+(I+1.4)
លំពែច		CALL SYMENTIE S. 1.15 .A
0110 0110	55.0	
0110 011.	2.00	- CONTRACTOR AND A REAL
0111 		- CREE STORY 14.015.01.1111.111.401.1420 - DO ECO 1.40 DITC
0112 6445		10 360 1900F110
0113 3114		ala=b.b+l+3. Pill.
9114 Sec.	-	UHLE SYME(14.5+71+.1+13.+90.+-2)
0115	26.0	CONTINUE
ចុ116		UBEL HUMBAR, 375+3, 3+, 1+50, +0, +-1+
0117		CALL NUMBER 4.3.55.1.0901.
• 11		I05 570 I=1+4
6119		VTIC=I+.6 F.CALE
£120-		WI=3.225++I+.6+
0121		CALL NUMER, 4.VI., 1.VIIC.90V.
012.	570	CONTINUE
	-	

PROGRAM DS2 (CONT)

		TV: 1 (±10
12422 2442		11/12/12/12/11
0124		111、12、12、12(11) オペンコンモア国際者
0120		THERE AND A THE AND A
0120		19767-2000 19767-2000
012		1997-2097 1997-2097
6128		11/07-2017
0129		11577年1月2日 11777年1月1日
0130		1:102744800
0131		SIN ZZEGRU TYZNANA OUTO
0132		ттали смырл э.э.с. тэ.түстүсөй стэ
0133		UMUL CHNDN,290,79,12711847779,717 Cont Diot, F.S. F.S.
0134		UHLE FLUIX.090.0900 To Foo Int O
0135		- 10 000 1-100 - 10 000 1-100
0136		АТЕВ.ОТКІЛ.2027 Соль сумр. Б.МТ. 1.10.200 .=00
0137		UHLL STABA.D)))),,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,
0138	580	LONTINOE Do roo I-O to
0139		10 590 I=0:10 11 - E. (1.1.1.)
0140		NIF.3+(1+1.4/
0141		CHLL SYMBUX199.00.1913.00.9747
0142	590	CONTINUE
0143		10 600 (=0,8
0144		YI#9.5-(1+.3/3)
0145		CALL SYMB(14.5) (1.1) 13.) 90. (-2)
0146	600	CONTINUE
0147		IY(1)=14
0148		IY(2)=2HBH
0149		IY(3)=2HLB
0150		IY(4)=2HNC
0151		IY(5/=2HE
0152		IY(6)≠2HAN
0153		17(7)=2HGL
0154		IY(⊗)≠2HE
0155		CALL NUMBA7.375.6.3.1.5001)
0156		CALL NUMB(.4+6.55+.1+0.+90.+-1)
0157		D0 610 I=1+4
0150		YTIC=90+I
0155		YI=6.38++I+.75+
0160		CALL NUMB(.4.711.7TIC:901)
0161	610	CONTINUE
0162		CALL SYMB(.2.7.3).12)IY(1),90.(1)
0163		CALL PLOTE.5.5.407
0164	0	THE REMAINING CALLS FOR PLOTTING DATA WILL GO HERE
9165		DO 730 I=1,3
0166	700	COUNTED
0167		ISTAT=1
		CALL PLOT(.14, VHT(1), 3)
Q1-99	710	EX=ECC(ISTAT)+.0015
0170		AM=EM+EM+SIN()(I) EN)(100)
0171		ANNIN-EX+EX+SIN(T+) + EX+ 300
0172		IF_CCAX-AGMIN+ .GT. BALCISTAT++ GOTO 2.4
3173		IF (I .EQ. 1 · COUNT=1
0174		TCT(ISTAT,I)=TCT(I)AT,I)+1
0125		YHT(ISTAT)#YHT(ISTAT+++AU+6C+6S+ACE_EC+
4176		BHEVINTHIDEBBEVINTHID-AC
4177		ELU(15 H()=EU(15 H()+.00:
<u>1128</u>	720	X=191H1+.14
0179		UHLE PLUTTING AND ANTALANDA
0150		(5(H)=15(H)+1 15 (2)(2)(5) (1) (1) (1) (1) (1) (1)
0101		16 (1316) .L. 101 (5000 -10 16 (2000) - 21 (6) - 2000 -200
0100 Atom		те соонналата от Ботбано соотност
0150	130	CONTRACT OF A DI
N154		UNCL MEDICO.(0.)32

PROGRAM DS2 (CONT)

,

0185		WRITE(6,17)
0166		LINE=5
0187		DO 800 l=1.100
0188		LINE=LINE+1
0189		IF (LINE .11. 5) 6010 750
0190		NRITE(6,13)
0191		LINE=0
0192	750	NRITE(6,18) L.TOT(1,1), TOT(1,2), TOT(1,3), ATHETA(1,
6143		-REAR(I)*REAR(I)*PLOALE
0194		ATHETA LISATHETA LI LEN.
0195		BWGHT() = BWGHT() + CSCACE
0196		YDATA(I.I.)=RUCHT(I)
6197		YDATA(1.2)=FBAR(1)
ñ198		YDATA(1,3)=ATHETA(1)
0149	800	CONTINUE
0200		DO 1000 I±1.3
0201		CALL PLOTE 13. Whore 1
0202		BO 900 ISTOTAL 100
0202		DO 200 ISTATE 13 Vetetate 13
0200		07191114117 Péli Blát, 1.98676/16767 (s. 5
0204 0205	ទាពល	CONTINUE
លាក់លក់ សុខភាព	200	IF (I FO 1) CARL DECT.
0200		IF (I FO D. CONTROL - LOTAN
0200	1000	AN AN ADMA ZA COLE FEURIO,95,9757 Continue
0200	1000	CONTROL CALL HEITE
020		DETHEN
0210		E LORIA Exam
0211		2110 CMD+
0412		C11D4

7649-80

