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YPG REPORT NO. 8027

ENGINEERING TEST OF

GANTRY, LIGHTWEIGHT, AIRDROF RIGGING

FINAL REPORT

BY

TERRY W. PUCKETT, 11T NOVEMBER 1968

STAIR CONTRACTOR CONT

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### DEPARTMENT OF THE ARMY Headquarters, U.S. Army Test and Evaluation Command Aberdeen Proving Ground, Maryland 21005

#### AMSTE-BG

#### 30 December 1969

SUBJECT: Final Report of Engineering and Service Test of Gantry, Lightweight, Airdrop Rigging, USATECOM Project Ncs. 4-5-7491-05 and 4-ES-655-035-001

Commanding General US Army Materiel Command ATTN: AMCRD-FS Washington, D. C. 20315

#### 1. References:

a. Preliminary Report of Engineering Design Test of Modified Gantry, Lightweight, Airdrop Rigging, DA Project No. 1F141812D183, Task 22.

b. Letter, AMSTE-BG, USATECOM, 20 December 1968, subject: Engineering Test Report of Gantry, Lightweight, Airdrop Rigging, RDT&E Project No. 1M141812D18322A, USATECOM Project No. 4-5-7491-05.

2. <u>Approval Statement</u>: Subject reports are approved except as stated herein.

3. Background:

a. Presently field units employ several types of materiel handling equipment not specifically designed for lifting loads being rigged for airdrop. To provide a single standard item, US Army Natick Laboratories developed the subject gantry system.

b. The complete gantry system has a lifting capacity of 35,000 pounds and consists of four "A" frame structures, two power packs and four accessory beams. This system can also be used as two separate lifting devices with each having a capacity of 17,500 pounds and consisting of two "A" frames, one power pack and two accessory beams. The gantries have an internal vertical clearance of 14 feet and a horizontal clearance of 12 feet. These clearances enable the gantry system to lift a nine foot high load five feet for placement onto transport vehicles ten feet wide. AMSTE-BG 30 Dec 1969 SUBJECT: Final Report of Engineering and Service Test of Gantry, Lightweight, Airdrop Rigging, USATECOM Project Nos. 4-5-7491-05 and 4-ES-655-035-001 Į.

Each gantry is provided with two suspended lift hooks of adjustable height to accommodate the various locations of lift fittings on various type cargoes. These gantries are equipped with caster wheels, incorporating a locking and unlocking device to allow manual positioning at the rigging site or for relocating the gantries. Screwtype legs (feet) of sufficient size are provided on each gantry to enable lifting of loads on soft unprepared ground or on hard snow. The total weight of the 35,000pound system is approximately 7,500 pounds and can be disassembled into component parts to permit manhandling by four personnel. The disassembled system is suitable for transport in air or ground vehicles.

c. The engineer design test of the gantry was completed in May 1966. At that time engineering tests (ET) were waived and the system was submitted for service testing. The results of this service test (ST) indicated that the gantry system was not suitable for Army use. As a result of the ST In-Process Review in December 1967, it was determined that the gantry system would be modified and would undergo both engineering and service tests.

d. The ET of the modified gantry was completed by Yuma Proving Ground in November 1968, the ST was completed by the US Army Airborne, Electronics and Special Warfare Board in September 1969 and the arctic winter service test is scheduled to be conducted beginning in September 1971.

4. Test Results:

a. The gantry met 34 of 50 requirements of the SDR. Six deficiencies and ten shortcomings were reported by the test agencies on the modified gantry system. After analysis and appropriate reclassification, no deficiencies and 15 shortcomings remain. One requirement of the SDR, operation and storage at  $-65^{\circ}$ F, will be evaluated during the arctic winter service test.

b. Deficiencies - None.

c. Shortcomings (15)

(1) It was very difficult for four men to carry the power pack assembly and load it onto a military vehicle. However, if the hydraulic oil was drained from the power pack, the weight would be reduced by

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65 pounds and, at this reduced weight, the test item will meet the requirements of the SDR. Further, the draining of the hydraulic oil (No. 10 engine oil) is a simple operation.

(2) The draft technical manual 5-3950-205-14 for the gantry system was not clear, concise, or complete. Eighty-eight recommended changes to this manual were proposed, 80 are classified as administrative, seven recommend that an operation is within the operator's capability (now assigned to other than the gantry operator), and one recommends that gloves be worn when operating the gantry in the manual mode. All of the proposals can be incorporated into an updated manual without being verified by retesting, and this item can be maintained under field conditions.

(3) Auxiliary equipment required for lifting loads was not included with the test system. Since MB-2 tiedowns, FSN 1670-545-9063, were found to be adequate during this test program and are available within the supply system, the addition of MB-2 tiedowns to the test system is considered acceptable.

(4) Spare parts for the power pack engines were not available through normal supply channels at the service test agency. Since these 3-HP gasoline engines are military standard items with hydraulic pumps and control valves, the lack of spare parts within the supply system should not be considered as a failure of the test item to meet the technical specification.

(5) During the ST this system demonstrated a 90 percent reliability with a 90 percent confidence level of completing a daily mission. (Requirement -95 percent) However, prior to the ST (during ET) the same system demonstrated the necessary capability of 95 percent reliability. Seven of the reported failures cited in the ST report (Appendix IV, Table 1) are not failures as defined in the SDR. These reported failures did not prevent the test system from completing its assigned mission and could be repaired by the operator with the tools and materials provided within 30 minutes. Revised reliability computations, utilizing the data from the ET and the revised data from the ST, are provided as Inclosure 1. These combined results indicate that the test item demonstrated a reliability of 94 percent with a 90 percent confidence level. This difference (one percent) between the requirement (95 percent) and that demonstrated (94 percent) by the test item is considered minor.

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SUBJECT: Final Report of Engineering and Service Test of Gantry, Lightweight, Airdrop Rigging, USATECOM Project Nos. 4-5-7491-05 and 4-ES-655-035-001

(6) The remaining shortcomings were found in the areas of physical characteristics, maintenance, operational performance/technical characteristics and human factors.

d. The test agency stated (deficiency) that the test system/device is not capable of immediate effective employment. The revised approved technical characteristics stated that the assembly time for a device (17,500-pound capacity) must be less than one hour. This was demonstrated during the ST (actual 47 minutes). Although the parameter for assembly of the system was not defined in the revised technical characteristics, the assembly of the system did take 94 minutes.

e. The test item met the requirements for maintainability.

f. The test item is safe to operate; however, potential operational hazards are noted below in paragraph 4g.

g. During this test program the following improvements/actions are suggested relative to the test item.

(1) That hydraulic hose(s) or hydraulic fitting(s) not be repaired but replaced as necessary.

(2) The cable ends be soldered in lieu of being taped. Tape falls off the cable ends in a short period of time and the loose ends are exposed.

5. <u>Conclusion</u>: The gantry system is suitable for Army use under intermediate environmental conditions.

6. Recommendation: As many of the shortcomings as feasible be corrected.

FOR THE COMMANDER:

2 Incl 1. as 2. ST Report /s/ William H. Hubbard /t/ WILLIAM H. HUBBARD Colonel, GS Deputy Chief of Staff

AMSTE-BG 30 Dec 1969 SUBJECT: Final Report of Engineering and Service Test of Gantry, Lightweight, Airdrop Rigging, USATECOM Project Nos. 4-5-7491-05 and 4-ES-655-035-001

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### Reliability Computations for Engineering and Service Tests

- 1. BASIC DATA:
  - a. Lift Cycles: 4102
  - b. Number of Failures: 2
  - c. Mission Day: 50 cycles

# 2. Computation of Point Estimate of Reliability:

a. Computation of point estimate of Mean Time Between Failures (MTBF):

MTBF = <u>NT</u>

MTBF =  $\frac{4102}{2}$  = 2051

MTBF = 2051 cycles or 5.56 mission days, where:

NT

= Test duration in terms of lift cycles, and

- = Number of failures
- b. Computation of Reliability:

R(x)	-	-x MTBF
<b>P</b> (50)	_	$\frac{-50}{2051}$
R(50)	-	2.73
R(50)	=	-024 2.73
R	-	.976 or 98 percent is a point estimate of reliability where:
		R = Point estimate of reliability
		x == One mission day == 50 cycles
		- Natural log base - 2.73
MTBF	-	2051 point estimate of Mean Time Between Failures

Inclosure 1

# 3. Computation of Reliability with 90 percent Confidence Level Assumed:

•

a. Computation of MTBF:

MTBF	-	2 × <sub>T</sub>				
		x <sup>2</sup> ,2	2 +2			
MTBF	-	$\frac{2(4102)}{x^2.1,2}$	$\frac{2}{2}$	,		
		A,		-		
MTBF	=	8204	5			
MTBF	-	770 cy	Cles	where:		
		×T		Test Duration	-	41 2 cycles
			•	Number of Failur	es (2)	
		x <sup>2</sup>	-	Chi square facto	r (from	Table H-3b.

 $x^2$  = Chi square factor (from Table H-3b, AMCP 702-3) and is given by 100(1-)% = 90 percent confidence.

### b. Computation of Reliability:

R(x)	- <u>MTBF</u>
	<u>-50</u> 770
R(50)	= 2.73
R(50)	-0649 = 2.73
R(50)	<ul> <li>.937 or 94 percent reliability at a 90 percent confidence level where:</li> </ul>
	R = Reliability
	x = Mission Day = 50 cycles
	Natural log base = 2.73
MIBF	Mean Time Setween Failures = 770 cycles

4. Results:

a. The test system demonstrated a point estimate reliability of 98 percent.

b. The test system demonstrated a reliability of 94 percent with a confidence level of 90 percent.

# USATECOM PROJECT NO. 4-5-7491-05

ENGINEERING TEST OF

# GANTRY, LIGHTWEIGHT, AIRDROP RIGGING

TEST REPORT

BY

TERRY W. PUCKETT, 1LT NOVEMBER 1968

YUMA PROVING GROUND YUMA, ARIZONA

#### PRECEDING PAGE MLANE\_MOT FILMED

#### ABSTRACT

The ergineer test of the Gantry, Lightweight, Airdrop Rigging, was conducted by Yuma Proving Ground from 20 May 1968 to 30 August 1968.

The purpose of the test was to determine the suitability of the test gantry for service testing.

All testing was conducted under natural environmental conditions. The approved technical characteristics of the test item were used as criteria to determine test item reliability. The power pack was too heavy for four men to carry and load onto a military vehicle (deficiency). The manual chain hoists corroded, the winch broke, and the hydraulic cylinder leaked oil (shortcomings).

It was concluded that the shortcomings were readily correctible, that the Gantry, Lightweight, Airdrop Rigging, is suitable for lifting loads up to 17,500 pounds when used as a device and 35,000 pounds when used as a system. It was recommended that the Gantry, Lightweight, Airdrop Rigging be subjected to service testing.

# FOREWORD

Yuma Proving Ground was résponsible for test execution, and préparation of the test réport.

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#### SECTION 1. INTRODUCTION

#### 1.1 BACKGROUND

In rigging and preparing airdrop loads, field units presently must employ several types of materials handling equipment such as mobile warehouse and rough-terrain forklift vehicles, heavy ordnance wreckers, construction cranes, and field fabricated A-frame gantrier with manually operated hoists to accomplish necessary lifting. These devices are inadequate for rigging loads up to the required weight of 35,000 pounds and are not readily available for use at remote outloading sites.

To remedy this situation, an SDR was approved in July 1964 for a Gantry, Lightweight, Airdrop Rigging (GLAR) to provide a single item with a wide range lifting capability to replace the various materials handling equipment now in use (Ref 4, App VI).

A prototype gantry was designed and fabricated, and testing was initiated at U.S. Army Natick Laboratories (USANLABS) on 1 November 1965. The prototype gantry was unsatisfactory from the viewpoint of safety and human factors engineering.

Testing of a redesigned and modified gantry was resumed by USANLABS on 15 March 1966. This engineer design test was completed on 30 May 1966. At that time, it was recommended that the gantry be submitted for service testing (Ref 2, App VI).

Engineering tests on the GLAR were waived. The U.S. Army Airborne, Electronics and Special Warfare Board (USAAESWBD) conducted a service test of the GLAR at Fort Bragg, North Carolina, from 9 January through 20 April 1967. Because the GLAR did not meet several requirements of the SDR, the service test was terminated prior to completion. The USAAESWBD recommended that the Gantry, Lightweight, Airdrop Rigging be considered not suitable for Army use and that consideration be given to the development of a device employing a lifting means which would eliminate the excessive maintenance and training requirements.

As a result of a Pre=In-Process-Review Conference on 2 November 1967 and a formal In-Process-Review Meeting on 6 December 1967, it was decided that the gantry should be modified and submitted to USATECOM for conduct of engineering and service tests beginning 1 April 1968. Yuma Proving Ground was designated to conduct the engineering tests.

1.2 DESCRIPTION OF MATERIEL

The gantry device (two gantries, Fig. 1) has a lifting capacity of 17,500 pounds and consists of two A-frame structures and a power pack. The gantry system (four gantries, Fig. 2) can provide a lift capability of 35,000 pounds. Each gantry has a clearance of 14 feet vertically and 12 feet horizontally, to allow lifting 9-foot-high loads up to 5 feet for placement onto transport vehicles 10 feet wide.



FIGURE 1. Two gantry devices lifting separate loads.



FIGURE 2. C.: gantry system lowering 35,000-pound load onto aircraft loader.

The total weight of the 17,500-pound capacity gantry device, suitable for both mechanical and manual operation, is approximately 3840 pounds. Disassembly into component parts is possible to permit manhandling.

Each power pack consists of the standard military gasoline engine with hydraulic pump and control and safety valves. Valves are arranged so that the gantries may be operated individually or in pairs. Lifting is achieved by a combination of two hydraulic cylinders with a pulley and cable arrangement on each gantry.

Gantries are equipped with caster wheels to allow manual positioning at rigging sites. Adjustable leveling feet are provided for support during lifting. Screw-type legs are raised and lowered by reversible ratchet wrenches on the gantries. Feet are of sufficient size to enable lifting on soft, unprepared ground surfaces.

Hand operated winches are provided for manual erection of the gantries. Also, manually operated chain hoists are included for lifting of the load in the event of power failure or as the only means of lifting.

Components of the disassembled system are suitable for transport in Army ground vehicles and aircraft.

1.3 OBJECTIVE

To determine the technical performance and safety characteristics of the gantry in accordance with the SDR and the advanced data package.

1.4 SUMMARY OF RESULTS

a. The Gantry, Lightweight, Airdrop Rigging, had the following deficiency and shortcomings:

(1) Deficiency. The power pack was too heavy for four men to carry and to load onto a military vehicle (Para. 2.2.3 and App III).

(2) Shortcoming. The manual chain hoist chains corroded while exposed to the desert environment (Para. 2.2.3 and App III).

(3) Shortcoming. A wrench lever would not lock and had to be held in position so that the baseplate could be lowered (Para. 2.3.<sup>h</sup> and App III).

(4) Shortcoming. The hydraulic oil cylinders were seeping oil which could not be stopped (Para. 2.4.3 and App III).

b. The following safety hazards, not covered in the safety criteria established by the USAAESWBD, were encountered:

(1) Cable ends of the winch hoist cable are clamped and the loose ends taped. After 3 or 4 weeks the tape falls off and the wire cable is exposed.

(2) Lifting loads with the manual chain hoist for 1 hour will cause blisters and open wounds on a man's hands (Para. 2.3.4d and App III).

(3) Shipping tape can become wedged between the winch hoist handle washer and the winch hoist causing the safety brake to slip (App III).

(4) When the angle from the accessory beam center fitting to the load connection point exceeds 3.6 degrees it is possible for the cable to jump out of the sheave when lifting the load. The load will drop approximately 8 inches when this occurs (Para. 2.5.3b).

(5) If failure occurs within the area of the hydraulic hose and the fittings located between the flow control value and the cylinder, the load will fall. No ettempt should be made to repair the referenced hose or flare the flange portions of the swivel nuts (Para. 2.5.3 and 2.5.4).

#### 1.5 CONCLUSIONS

8

a. The above deficiencies and shortcomings are readily correctible.

b. The Gantry, Lightweight, Airdrop Rigging, is suitable for lifting loads up to 17,500 pounds when used as a device and 35,000 pounds when used as a system.

#### 1.6 RECOMMENDATIONS

The Gantry, Lightweight, Airdrop Rigging be subjected to service \*esting.

#### 2.1 INTRODUCTION

The Gantry, Lightweight, Airdrop Rigging, hereafter referred to as the test item, was tested by Yuma Proving Ground during May through August 1968.

The Approved Technical Characteristics of the test item were used as criteria to determine test item reliability.

The requirements that the system be capable of statically supporting twice the rated load without evidence of permanent deformation and that the system demonstrate sufficient reliability and durability to lift 150 percent of its rated load to a height of 60 inches for 50 cycles were deleted from the test (Ref 5, App VI).

Testing to determine if test item met criteria listed in Paragraph 2.3.1.1c was not conducted due to satisfactory completion of the test during the USAAESWBD service test conducted in June 1967.

2.2 TEST NO. 1 - PHYSICAL CHARACTERISTICS

#### 2.2.1 Objective

To determine the physical characteristics of the test item.

2.2.1.1 Test Criteria.

a. Individual components of the system shall be sufficiently lightweight to enable carrying for short distances and loading by four men onto a military vehicle (Para. 1, App II).

b. No component or group of components of the system shall be of such a size as to prevent air transport by cargo aircraft in accordance with applicable portions of Appendices A and B of AR 705-35. Component parts of the system must comply with the requirements of approved specifications (federal, military, and/or industry), and be made corrosion resistant through use of applicable methods and materials (Para. 3, App II).

#### 2.2.2 Method

The test item components were examined, measured, weighed, carried 25 yards by four men and loaded onto a military vehicle. Technical characteristics were reviewed and checked against AR 705-35. A weight comparison was made between a test item having a hydraulic system and a test item having a manual chain hoist.

#### 2.2.3 Results

a. Weights of the test item and its components are contained in Table 1, Appendix I.

b. No components or group of components of the test item exceeded 15 feet in Length, 70 inches in width, and 60 inches in height.

c. Individual components of the system (as listed in Table 1, App I), with the exception of the power pack assembly with hydraulic oil, were carried 25 yards by four men, and loaded onto an M35A1, 2-1/2 ton truck. The power pack assembly could be carried with extreme effort but could not be loaded onto the M35A1 truck.

d. The chains on the manual chain hoist rusted while exposed to the desert environment.

# 2.2.4 Analysis

All components of the test item are within the size limitations for air transport by cargo aircraft.

The power pack assembly with hydraulic oil cannot be carried and loaded onto a military vehicle by four men without complete disregard for the safety and physical well being of the men involved. However, if the hydraulic oil were drained from the power pack the weight would be reduced by approximately 65 pounds; at this reduced weight the test item can be carried by four men.

After exposure to the climatic conditions of the test site for 3 weeks, rust began to appear on the manual chain hoist chains. This occurred under no rain, low humidity conditions.

2.3 TEST NO. 2 - OPERATI')NAL TEST

2.3.1 Objective

To determine the operational suitability of the test item.

2.3.1.1 Test Criteria.

a. The system must be cabable of manual assembly, from shipping to operational condition, without special tools or materials handling equipment. Assembly time for a device (17,500-pound capacity) from removal from shipping skids to erection must be less than 1 hour, when using four men.

b. The device must have a lifting capabily of 17,500 pounds and when used in pairs as a system must be capable of lifting a load measuring 108 inches high, 110 inches wide, and 336 inches long, weighing 35,000 pounds, to a height which will provide a 60-inch ground clearance and will permit placement onto a ground transport vehicle up to 120 inches in width.

c. The system must have mechanical leveling provisions to insure stability in all directions for all loads up to rated load on sloping terrain up to and including 5-degree slopes.

d. The device, when assembled, must be capable of being man-propelled short distances over unsurfaced and non-trafficked areas in the vicinity of forward airfields.

e. The system must be capable of raising the rated load to a 60-inch height in approximately 120 seconds, using self-contained gasoline engine operated power packages, together with hydraulic control and lift components.

f. The device shall be operable from a single control station.

g. The system must be capable of manual operation if power is not available. With manual operation, the lift rate requirement listed for mechanical operation is not mandatory.

h. The system must meet the requirements of the current revision of Specification MIL-T-11748 (Signal Corps), "Interference Reduction for Electrical and Electronic Equipment."

2.3.2 Method

a. Time required for manual assembly of the test item from shipping to operational conditions was recorded. Four men, one NCO who was given on-the-job training for approximately 4 hours, and three enlisted men who were given a 15-minute briefing on the erection of the gantry device, assembled the test item.

b. The test items were used for a period of 3 months to lift various loads within the stated weight and dimensional limitations. Lift and lowering times were recorded.

c. The test item was man-moved for short distances when assembled and disassembled. It was man-moved when assembled as both a manual and a mechanical device. The areas in which these tests were conducted were composed of rock alluvium, the dominant features of gravelly deserts, the most common desert type, made up of gravel stratum mixed with sands and silts.

d. The test item was tested manually and mechanically and the regults were recorded.

e. Manpower necessary to operate control stations was observed.

f. A radio interference test was conducted on the test item and the results recorded. Specifications MIL-STD-461 and 462 were used in place of MIL-T-11748 as it was not available during testing.

g. Still pictures were taken and analyzed.

h. Engineering data were recorded as necessary.

- Contraction

i. A test load weighing 35,000 pounds was modified to simulate a load measuring 110 inches wide and 336 inches long. Height measurements were taken to determine height limitations. A ground transport vehicle (aircraft loader) measuring 120 inches in width was then driven between the gantries to permit load placement.

2.3.3 Results

a. The gantry device was manually assembled from shipping to operational condition, without special tools or materials handling equipment, in 58 minutes.

b. Operational data are contained in Table 2, Appendix I.

c. The test item was man-moved a distance of 300 feet when assembled for manual operation and when assembled for mechanical operation.

d. The test item was manually operated with loads up to 17,500 pounds using one device. A 35,000-pound system could not be tested due to the nonavailability of enough chain hoist during testing. Operational data are contained in Table 3, Appendix I.

e. One control station is required when operating a gantry device. Two control stations are required when operating a gantry system.

f. The test item met the "Interference Production for Electrical and Electronic Equipment" requirements. Data are contained in Table 4, Appendix I.

g. A ground transport vehicle up to 120 inches wide can be driven between a gantry system and pick up a load which is 110 inches wide, 336 inches long, and 108 inches high (Fig. 1, App V).

### 2.3.4 Analysis

Criteria were met with the following exceptions:

a. During erection of the test item, a wrench lever would not lock and had to be held in position so that the baseplate could be lowered.

b. Although the test item was assembled from shipping to operational condition by four men in less than 1 hour, it must be noted that the cart assembly was received with the cylinders and hydraulic hose connected. Had the cylinders been shipped as separate units in the shipping containers constructed by the U.S. Army Natick Laboratories, the time

requirement would not have been met. Also, both ends, as well as the top of the shipping container must be removed if no materials handling equipment is utilized. (Future shipping procedures should be determined.) Since the test item was shipped from Natick, Massachusetts, to Yuma, Arizona, with the cylinders mounted and no damage was incurred, it is our opinion that special cylinder crates are unnecessary. It was also noted that the shipping skids used were nailed down in a kite box shaped crate. If the shipping skids were constructed in the shape of a solid rectangular box with a fold back lid, the time required to remove the test item from the shipping skids would be considerably less (Ref 3.5, App III).

c. The gantry was moved a distance of 300 feet in 1 minute and 50 seconds without difficulty when assembled for manual operation. When assembled for mechanical operation the gantry became stuck in the sand on two occasions (movement time 2 minutes and 20 seconds) and although movement was continued with added effort, extreme caution had to be taken to prevent the gantry from toppling. It is also noted that since there are numerous types of soil groups in forward airfield areas, only one of which was available at this testing area, satisfaction of the requirement cannot be determined.

d. Although the test item may be manually operated with loads up to 35,000 pounds, the physical stress on *m* man's hands creates numerous blisters and open wounds (Fig. 2, App V). Therefore, gloves to eliminate this situation should be standard issue with test item.

2.4 TEST NO. 3 - MAINTENANCE AND RELIABILITY

2.4.1 Objective

To determine if the test item meets maintenance and reliability requirements as defined by the Operational and Technical Characteristics.

2.4.1.1 Test Criteria:

a. The system, when operated by its hydraulic power package, shall demonstrate with 95 percent reliability the capability of performing a daily mission. A daily mission is defined as a total of 50 cycles (lirts of various load weights within the rated capacity. This implies 20 mission days as Mean Time Between Failures (MTBF). A failure is defined as that which prevents the unit from completing its assigned mission and crunot be repaired by the operator with the tools and materials provided within 30 minutes. Unscheduled organizational maintenance should not exceed 30 minutes during the performance of a daily mission. The total maintenance manhours will not exceed 10 percent of the operating hours on the basis of 8 hours of operation equal to 1 mission day. Total maintenance will include scheduled and unscheduled maintenance from operator level through direct support level.

b. The system must be easily maintained under field conditions. Components must be interchangeable between like items of the system. Maintenance costs must be a minimum for systems of this type.

c. The system must be capable of operation and storage in temperatures from  $-65^{\circ}F$  to  $+125^{\circ}F$ .

### 2.4.2 Method

a. An updated Draft Technical Manual, Manufacturers' Maintenance and Operating Manual, and TM 5-2805-203-14 were the only guides available in performing all maintenance during the conduct of this test.

b. A preoperational inspection was performed in accordance with pretesting procedures. However, preoperational inspection time was not recorded as it is not considered a portion of maintenance time. Technical inspections were conducted by maintenance personnel as required. Daily inspections and preventive maintenance operations were performed as directed in Draft Technical Manual and TM 5-2805-203-14.

c. Unscheduled and scheduled maintenance was performed as required. Records were maintained for all maintenance operations to include time required and reasons for actions performed.

d. Soil and ambient air temperatures and relative humidity were recorded during the operation of the test item.

e. Four individual test items were used as a device (two gantries used in unison) and as a system (four gantries used in unison). Each individual gantry was operated a total of 50 cycles per day for 20 working days.

#### 2.4.3 Results

a. The test items, four each, which were used as two devices and as one system during testing, were operated for a total of 91.2 engine hours. During these operations, 1.7 manhours of scheduled and unscheduled maintenance were performed. The scheduled maintenance consisted of 0.5 menhours for the 25-hour organizational maintenance of the gasoline engine.

For details of scheduled and unscheduled maintenance, see Table 1, Appendix IV. In addition, operator daily inspection and servicing were performed requiring approximately 5 minutes per day.

Daily preventive maintenance was performed by test personnel without difficulty.

b. Eleven unscheduled maintenance actions occurred during engineer testing, requiring a total of 1.2 manhours to accomplish (Table 1, App IV).

c. Table 2, Appendix IV, shows gantry operating hours, active maintenance time, maintenance rates, and mean time between railures. The maintenance ratio, based on scheduled and unscheduled maintenance actions but exclusive of operator daily preventive maintenance, and initial inspection was 0.019 for the entire engineering test period.

d. A limited amount of repair parts were furnished with the test item; however, most replacement parts were either locally obtained or fabricated. Parts which could not be obtained in this manner were taken from like items of the system. Components are interchangeable.

e. Some of the unscheduled maintenance was of a serious nature and required a high degree of maintenance skill. However, the test NCO, who carries an MOS of 62E40 (Heavy Equipment Supervisor), and other test personnel were able to correct all unscheduled maintenance without outside assistance with the exception of oil leaks which came from the cylinder seals and could not be corrected.

f. Soil temperatures ranged from  $73^{\circ}$ F to  $109^{\circ}$ F. Ambient air temperature ranged from  $68^{\circ}$ F to  $107^{\circ}$ F. Relative humidity ranged from 17 to 64 percent. For daily meteorological readings refer to Table 2, Appendix I.

2.4.4 Analysis

a. The amount of maintenance required and the number of parts consumed during engineering testing were within stated criteria.

b. The system, when operated by its hydraulic power package, will perform a daily mission with 95 percent reliability.

c. One failure, Item L, Table 1, Appendix IV occurred due to attempted repair. It has been established that this item should be replaced and repair should not be attempted for safety reasons. Replacement time is 15 minutes.

d. Oil seepage from the cylinder seals (Fig. 3, App V) could not be stopped by tightening. Although oil seepage was very slight, it could cause damage to the parachutes of a rigged load. The seals should be replaced with a higher grade seal to avoid leakage.

e. Since the gantry was tested under desert environmental conditions for over 20 days of operation, it can be assumed that the test item will operate under extreme high temperatures. Although ambient temperatures did not reach  $125^{\circ}F$  the system is capable of operation under such conditions.

Test item operation at low ambient temperatures could not be determined at this test site.

f. Pressure gage failures (four each) were not considered either a deficiency or shortcoming; however, correction of these failures required 15 minutes of active maintenance time. Of the four failed gages, three were furnished by this installation and had brass movements. One failed gage, which was an initial component of the test item, contained a bronze movement (Para. 3.1, App III, and Fig. 4, App V).

2.5 TEST NO. 4 - HUMAN FACTORS AND SAFETY

2.5.1 Objectives

To determine if the test item conforms to the principles of human factors engineering.

2.5.1.1 Test Criteria. The test item must be safe for its intended use.

2.5.2 Method

The safety criteria as established by the USAAESWBD after service testing in June 1967 was reviewed and followed. Any safety hazards which were not discovered during the service test were recorded.

2.5.3 Results

During the 20-day reliability test of the gantry system, one device had performed the following:

50 lifts of 12,000 pounds each 40 lifts of 12,500 pounds each

Total operating time was 3 hours and 25 minutes. After 67 lifts a small oil leak was noticed which was coming from Items A and B of Figure 3.

The leak was very small at the time (3 or 4 drops of oil per lift). Testing was continued and the leakage kept under observation. At the completion of 90 lifts the leak had increased (10 to 12 drops of oil per lift) and operation was ceased. An attempt was made to tighten the left bolt (Item A). When this was done the separation around the flange (Item B) increased due to the tautness of the manifold line (Item C, Fig. 3).

The entire assembly (Items A through D, Fig. 3) was then removed. Teflon tape was placed around Item A, Figure 3, and a flaring kit was used to flare the flange out on Item B, Figure 3. The assembly was then reinstalled. The motor was started and the load lifted to approximately 36 inches off the ground. The position valves were then pushed down so that the load would begin to descend.

At this time the flange "blew out" of its fitting, spraying oil approximately 30 feet in both directions, and the load suspended by the gantry dropped to the ground.

Further investigation disclosed that the clamp connections (Items E and F, Fig. 4) failed to restrain the manifold hose in its proper location. The hydraulic "hammer action" over the period of 3-1/2 hours of operation and 90 lifts pulled the manifold hose in a direction away from the fitting and through the clamps approximately 1 inch. This,

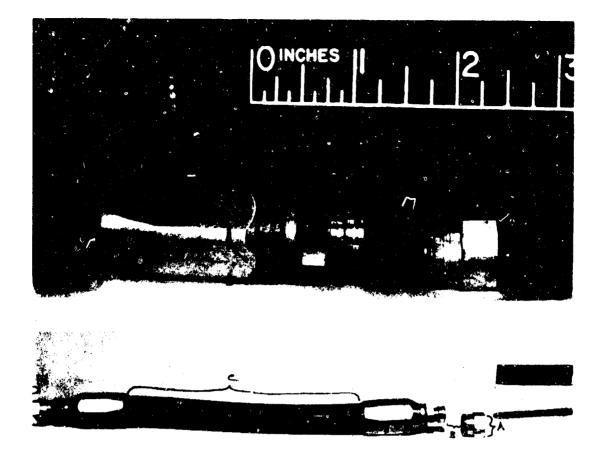


FIGURE 3. Hydraulic hose and fittings.

in turn, pulled the hose so taut that all force was being exerted on the final connection (Items A and B, Fig. 3). The continuing "hammer action" could have been the reason the fittings loosened and the leakage started. Hydraulic oil temperature at this point is 165°F with a maximum oil pressure of 2300 psi.

b. When the angle from the accessory beam center fitting to the load connection point exceeds 3.6 degrees it is possible for the cable to jump out of the sheave when lifting the load. The load will drop approximately 8 inches when this occurs (Fig. 5, App V).

c. Safety hazards (Para. 3.2 and 3.3, App III) should be corrected (Fig. 2 and 6, App V).

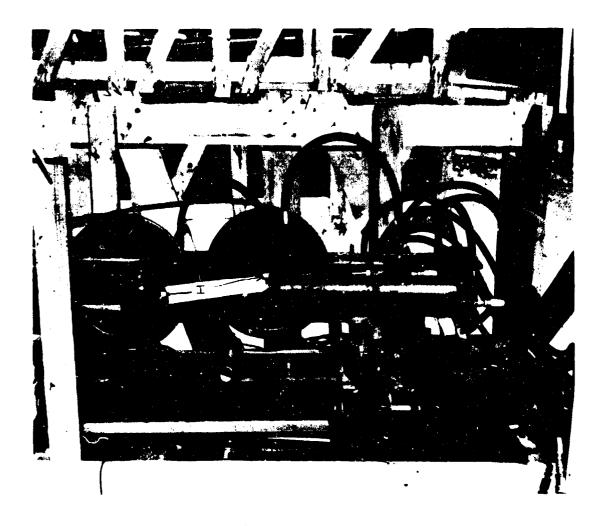


FIGURE 4. Cart assembly.

# 2.5.4 Analysis

a. The clamp connections, Items E and F (Fig. 4) should be checked daily to insure that they secure and that the manifold hose is restrained in its proper location.

b. The load will fall if the referenced hose assembly or fittings fail. These items are located between the flow control valve and cylinder, and should be checked daily for oil leaks. No attempt should be made to repair referenced hose or to flare the flanged portions of the swivel nuts. Operation should be ceased and the faulty items replaced.

c. Caution should be used during rigging to avoid connecting the gantry to the load at more than a 3.6-degree angle as this can damage the gantry cable and presents a possible safety hazard.

NOTE: Figure 4 is not a photograph of the gantry on which the incident occurred and is for reference only. Clamp E is normally located at Point G and Clamp F at Point H. Point I is the normal location for the hose shown in Figure 1.

# TABLE 1. Gantry Components

	Unit	Unit	Number	Number of Units					
Nomenclature of Component	of <u>Issue</u>	Wt ( <u>}b</u> )	Per Device (2 gantries)	•					
Left hand column assembly with winch	Fa	190	4	8					
Right hand column assembly with stay bar	Ea	170	4	8					
Main beam assembly with hinge pins and rods	Ea	290	2	24					
Cart assembly with cylinders and hydraulic hose	Ea	365	2	4					
Accessory beam	Ea	100	2	4					
Power pack assembly with hydraulic oil	Ea	421	1	2					
Manual chain hoist	Ea	93.	5 4	8					
Tie-down chain assembly	Ea	10.	5 4	8					
5-gallon jerry can with gas and gas line	Ea	53.	5 1	2					

Total weight of gantry device (2 each gantries) for manual operation only: 2636.0 pounds.

Total weight of gantry device (2 each gantries) for mechanical operation only: 3478.5 pounds.

Total weight of gantry device (2 each gantries) for manual and mechanical operation: 3840.5 pounds.

TABLE 2. Horizonical Convention Data

Pressure Teristica Doma Idle

1200 1000 2000

Gestry Bo. 1 and 2 Losd Weight: 8060 lb Muriums Beight Lors Lifted: 52 is. Power Pack Bo. 1 Abliest Compersurve (\*P): Eigh, 75; Low, 75

1.1

Santry Bo. 3 and b Lond beight: 12,000 2b Marinam Feight Lond Lifted: 62 in. Power Fack Wo. 2 Ambient Temperature (\*Y): High, T0, Low, Tb

0.

Gentry Bo. 1 and 2 Lond Wright: 7500 1h
Haring Seight Lond Lifted: 52 Au.
Power Pack No. 1
Auhiest Temperstare (*7):
Eigh, Th; Low, Th
Bunidity (\$): Migh. 55; Low, 55

(sec) Down

0

1200 1900

<u>7188</u> 54

Cycle Bo.

Presentation Periodica Dorn Idle

1300 1300 1500

1100

1700

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3 5 6 T 8 9 10	41	45				3 4 5 6 7 4 9 10 11 11 14 15 16 11 14 19 20	47	52		
•	41	45				•	47	51		
5	1	*6				5	47	51 51		
6	-2	46				•	41 47	51		
Ī	•••	47			• • •		4 <u>7</u>	51		
8	2	11 10 11	119C		1900	ŧ	47 47 47	51		
9	41					, ,		51		
10	<u> </u>	•7				10	47	51		
14	43	47				<u>.</u>	•	51		
13	<u>مت</u>	4.9 4.2					4. 4. 4.	14		
6.6	4-2							51		
1	- <b>4</b> .	48 48					4. 42	- <del>2</del>		
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25	42	<u><u><u></u></u></u>				25	45	54		
20	۰.	21 22				25	45 45	50		
27	41	52				27	45	55		
28	-	52				23	45	- 55		
25	42	******				25	45	54		
3Ċ	•1	54				30	45	54		
3:	40	51				<u>):</u>	40	55 55 54		
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33	<b>4</b> 0	52				33	40	54		
خز	e.,	52				34	40	55		
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36	43	52 51				<u>)</u>	42	55 55 47 42		
- 31		- 51				37	42	-		
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*3								<u>, </u>		
• <u>1</u>	10	52						24		
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*C	40 40	2- 51				••• •*	43	52		
	40	21 41				-	1			
		51 52					47	÷.		
~	434	<u>x</u>						51 51 49		
~	~	~				-		.,		

Semarks: Dycle No. 42. Home connectors were disconnected from the power pack assembly and reversed so that the four-

assessive and reversed so that the four-way control value levers would lift and lower in conjunction. Upon replacing the connectors a heavy leak started in connection Sc. 3. No visual dwange to the intrior positions of the connectors was noted and leakage was stupped by trial and error adjustment of the hose fitting (EFR 15-2).

\$ \* let. ę 399533333533 1130 Agential After 5 minutes of test life in operation ace of the manifold nose con-nectors which was connected to the power pack began to leak. The fitting was tightened and hele for around 5 minutes. The fitting began leaking again and although tightened to 'ts capacity the oil leakage increased and neweloped into a heavy spray. Operation was ceased and the connector was disconnected for repair. The connector had not been filled with 'white lead" during production and was repaired by placing terion tape between the fittings [EFR 15-4). Cycle Xo. 15. Namifold bose 'jusped' out of the manifold hose support (EFR 15-5). mrke : No problems were incurred.

TABLE 2. Mechanical Operation Data (Continued)

Gentry Ho. 3 and b Lond Weight: 12,500 lb Haximum Height Lond Lifted: 62 in. Power Pack Ho. 2 Ambient Tampersture (\*F): High, 75; Low, 7b Humidity (\$): High, 55; Low, 55

Cycle Time (sec) No. Up Down

50998999779787877777799887718077

Not Reco

ded

orded

125456789011211415611828228288788881231455418690124344444444444

.

Pressure Variation Up Down Idle

1200 1000 2000

900

1800

1100

Load Maxiw Power Ambien High Humid	Weight um Hei Pack n Tem , 78; it, (\$	peratur Lov, 72 ): High	16 4 Lift • (*F) ., 21;	: Low, 1	8	Gantry No. 3 and 4 Load Weight: 13,000 lb Maximum Height Load Lifted: 62 in. Power Pack No. 2 Ambient Temperature (°P): High, 78; Low, 72 Humidity (\$): High, 21; Low, 18 Soil Temperature (°P): High, 75; Low,									
Cycle	Time	(sec)		vesure		Cycle	Time	(sec)		Pressure Variation					
lo.	Up	Down	Up	Down	Idle	No.	Up	Down	Up	Down	Idle				
1	43	47	1200	1000	1800	1	53	47	1200	<b>0900</b>	2100				
2	42 42	46 48				2	51	48							
3	42	48				3	51 49	49 48							
5	41	49	1000			5	50	49	1100						
6	41	49				6	50	49							
7	-1	49				7	51	49							
8	41	49				8	50	49							
9 10	41 41	50 50		1100		9	50	50							
n	<b>b1</b>	49				10 11	48 49	50 50							
12	41	١ĝ				12	48	50							
13	40	49				13	49	51							
14	41	49				14	49	50							
15	41 40	49 10				15	48	50							
16 17	40	49 50				16 17	49 49	50							
18	40	49				18	49	51 51							
19	40	49				19	48	51			2000				
20	41	49			1700	20	48	51			2000				
21	41	49				21	48	51							
22	40	49				22	48	51							
23	40	50				23	48	51							
24 25	40 40	50				24	49	51							
26	40	50 50				25 26 N	48 ot Rec	51 51							
27	4C	50				27	OL NEL	orueu							
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23	40	50				29									
30	40	50				30					1900				
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42	10	50				41 42	48	51 51							
43	40	51				43	48	51							
44	40	50				44	49	52							
45	40	50				45	50	52							
46	40	50				46	50	52							
47	40	51			1700	47	50	52			2000				
48 49	40 40	50 50			1700	48 49	50 48	51 52							
50	40	50				49 50	40	52							
						,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	-,	12							
Resau	rks: 1	No prob	lens v	ere in	curred.	Remar	ks: N	io prob	lems v	ere in	curred.				

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Remarks: Manifold hose "jumped" out of the manifold hose support approximately 50 percent of the cycles when the load was starting to be lowered from its suspended pesision (EPR L5-5-2). Cycle No. 17. Oil leak at connection of cylinder and hydraulic hose (EPR L5-6).

TAME 2. Mechanical Operation Data (Continued)

 Gentry Ho. 1 and 2
 Gentry Ho. 3 and 4
 Gentry No. 1 and 2

 Load Weight: 6560 lb
 Load Weight: 13,500 lb
 Load Weight: 6060 lb

 Maximum Height Load Lifted: 62 in.
 Haximum Height Load Lifted: 62 in.
 Naximum Height Load Lifted: 62 in.

 Power Pack No. 1
 Power Pack No. 2
 Power Pack No. 1

 Ambient Temperature (\*P):
 Ambient Temperature (\*P):
 Ambient Temperature (\*P):

 High, 75; Low, 74
 High, 75; Low, 74
 High, 74; Low, 68

 Humidity (\$): High, 21; Low, 17
 Humidity (\$): High, 21; Low, 17
 Humidity (\$): High, 40; Low, 20

 Soil Temperature (\*P): High, 81; Low, 77
 Soil Temperature (\*P): High, 81; Low, 77
 Loed Yeight: High, 82; Low, 73

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			F	ressur	4	Precaure							Pressure					
Cycle		(996)		le <u>ri et</u> i		Cycl		(999)		ristic			Cycle		• (sec)	V	ristic	<u>a</u>
<u>Bo</u> .	<u>Up</u>	Down	<u>Up</u>	<u>üteriti</u>	Idle	<u>Io</u> .	<u>5</u>	Down	<u>Un</u>	Down	Idle		Jo.	<u>Up</u>	Down	<u>Up</u>	Down	Idle
1	41	46	1200	1100	1700	1	49	47	1300	0900	1900		1	41	48	1100	1000	1700
2	41	46				2	50	46					2	40	49			
3	41	47		1000		3	59	49					3	40	48			
5	41 41	47 48		1000		5	50 50	47 48	1200				4 C	40 39	49 48			
6	41	48				6	50	48			2000		6	40	48			
7	41	48				7	50	48					7	39	48			
8	41	49				8	50	49					8	39	49	1000		
.9	40	50			1800	9	49	50					9	39	48			
10 11	Not Re	A COPDED				10 11	49 Not le	50					10 11	39 39	48 48			1800
12		1				12	MOL .4	A					12	39	48			
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19						19							19	39	49			
20						20							20	38	48			
21		1				21							21	37	50			
22						22		1					22	39	50			
23 24		l.				23 24	•	¥ i					23 24	39 40	50 49			
	Not R	v ecorded				25	Not Re	corded					25	38	51			
25 26	39	51	•			26	49	50					26	37	49			
27	40	50				27	50	50					27	39	só			
28	40	52				28	49	50					28	39	51			
29	41	49				29	49	48					29	38	50			
30	40 40	49 49				30 31	49 49	49 50					30 31	39 37	50 51			1700
31 32	40	50				32	49	50					32	40	50			
33	39	50				33	49	49					33	37	51			
34	40	49				34	49	50					34	38	51			
34 35 36	40	50				35	48	51					35	38	52			
36 37	41 40	51				36	49	50					36	37	50			
38	40	50 51				37 38	50 49	51 50					37 38	38 38	52 51			
39	40	50				39		corded					39	38	52			
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48						48		1					48	38	50			
49	N-4 P	۷				49		¥					49	38	51			
50	Not Re	corded				50	Not Re	corded					50	38	50			
Remax	ks: N	o probl	.em. ve	re inc	urred.	Rem	arks :	No prob	lems v	ere is	curred.	•	Renar	ika :	Но ргоб	lens v	ere in	curred.

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TABLE 2. Mechanical Operation Data (Continued)																				
Gantry Ho. 3 and 4 Load Weight: 14,000 lb Maximum Height Load Lifted: 62 in. Pover Pack Ho. 2 Ambient Temperature ("F): High, 84; Low, 68 Humidity (\$): Higb, 40; Low, 20 Soil Temperature ("F): High, 82; Low, 73						Centry Ho. 1 and 2 Load Weight: 5560 lb Maximum Height Load Lifted: 62 in. Power Fack Ho. 1 Ambient Temperature (°F): High, 96; Low, 93 Humidity (\$): High, 24; Low, 20 Soil Temperature (°F): High, 94; Low, 94						Gantry No. 3 and 4 Load Weight: 14,500 lb Maximum Height Load Lifted: 62 in. Power Pack No. 2 Ambient Temperature (°P): High, 96, Low, 93 Humidity (\$): High, 24; Low, 20 Soil Temperature (°F): High, 94; Low, 94								
Cycle <u>No.</u>	<u>Time</u> Up	(sec) Down		ressur Aristi Down			Cycle No.	<u>Time</u> Vp	(sec) Dova	V	vesure riatio Down			Cycle No.	<u>Tige</u> Up	(sec) Down	V.	tistic <u>Down</u>		
1	51	50	1200	1000	1900		1	36	49	1100	1100	1700		1	53	49		1000		
2	50	50					2	37	50					2	53	50	1200	1000	1000	
3	51	50 49					3	36	49					3	51	49				
5	50 50	49 51					4	36 37	49 50					4	52	49				
6	52	49					6	37	50					5	51 52	50 50				
7	52	1 A S					7	36	52					1	51	52				
8	52	51					8	36	51					8	52	51				
9 10	50 50	52 50					9 10	36 36	51 51					9	52	51				
n	51	51			2100		11	35	50					10 11	52 52	51 50				
12	51	50					12	36	50					12	52	50				
13 14	52 53	51 51					13	36	51					13	52	51				
15	51	52					14 15	35 36	49 49					14 15	52 52	49 49				
16	50	50					16	36	50					16	51	50				
17	52	52					17	.17	49					17	51	49				
18 19	52 51	51 52					18	36 36	50					18	50	50				
20	51	50					19 20	30 37	50 ຸງ					19 20	51 52	50 50				
21	52	51					21	37	50					21	52	50				
22	51	52					22	36	49					22	52	49				
23 24	51 50	5λ 52					23 24	37 37	49 49					23	51	49				
25	51	51					25	36 36	49					24 25	52 52	49 49				
26	52	52					26	36	50					26	51	50				
27 28	51	50					27	36	50			~		27	52	50				
20	52 51	51 50					28 29	36 36	50 50					28	51	50				
30	50	53			1900		30	37	50		1000			29 30	52 51	50 50	1400	1100		
31	51	48			-		31	37	50					31	51	50	1400	1100		
32	50 39	49 48					32	37	50					32	52	50				
33 34	50	40					33 34	37 37	50 50					33 34	51 52	50 50				
35	50	48					35	37	50					35	52	50				
36	50	48					36	36	50					36	51	50				
37 38	50 51	48 49					37 38	36 36	50					37	52	50				
39	50	48					30 39	30	50 50					38 39	51 51	50 50				
40	50	48					40	36	50					40	52	50				
41 42	50 50	48 48					41	36	50					41	52	50				
43	50 50	40					42 43	36 36	50 50					42 43	52 52	50 50				
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45	50	48					45	36	50					45	52	50				
46 47	j0 50	48 48					46 47	36	50					46	51	50				
48	51	49					47	36 36	50 50					47 48	52 52	50 50				
49	<b>50</b>	49					49	36	50					49	52	50				
50	50	48					50	36	50					50	51	50				

Remarks: No problems were incurred.

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Remarks: No problems were incurred.

Remarks: Cycle Ho. 13. Pressure gage broke. Item was replaced (Nef App III) (EPR L5-7).

TABLE 2. Mechanical Operation Data (Continued)

Centry No. 1 and 2 Load Weight: 5060 lb Maximum Meight Load Lifted: 62 in. Power Pack No. 1

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Gentry No. 3 and 4 Load Weight: 15,000 lb Maximum Height Load Lifted: 62 in. Power Pack No. 2 Gentry No. 1 and 2 Load Weight: 4950 1b Maximum Height Load Lifted: 62 in. Power Pack No. 1 823

	Pressure Time (sec) Variation		<b>0</b> 1				Pressure Variation C			Time (sec)		Pressure Variation				
Cycle No.		Dova	J.	Down	Idle	Cycle <u>Ho</u> ,		(aec) Dovra	Up Down		Cycle <u>No.</u>	Up Up	Down	Up	Down	141.0
1	38	50	1000	1100	1700	1	51	50	Not valid		1	36	49	1000	1200	1700
2	31	49				2	52	49			2	35	49			
3	37	49				3	52	49			3	35	49 49			
4	36 36	49 49				4 5	53 53	50 50			5	35 35	49			
6	36	49				6	52	50			6	32	49			
Ť	36	49				7	53	49			ĩ	35	49			
ė	36	19				ė	53	49			ė.	35	49			
ĝ	36	50				9	52	49			9	35	49			
10	35	49				10	53	49			10	35	49			
11	36	49				11	53	49			11	35	49			
12	35	50				12	52	50			12	35	49			
13	35	49				13	53	49			13	35	49			
14	35	50				14	53	50			14	35	49			
15	35	49				15 16	53 52	49 49			15 16	35	49			
16 17	35	49 49				10	53	49			17	35 35	49			
18	35 32	49				18	52	49			18	35	50			
19	35	49				19	52	iĝ			19	35	50			
20	35	50				20	53	50			20	35	50			
21	36	50				21	53	50			21	35	50			
22	35	4.7				22	52	49			22	35	£9			
23	35	49				23	52	49			23	35	50			
24	35	50				24	53	50			24	35	49			
25	25	50				25	52	50			25	35	49			
26	35	50				26	52	50			26	35	49			
27 28	35	1.5				27 28	52 52	49 49			27 28	35 35	49 49			
29	35 35	*9 50				20	53	50			29	35	50			
30	35	50				30	53	50			30	35	49			
ũ	36	50				31	52	50			31	35	49			
31 32	36	49				32	53	49			32	35	50	900	1100	
33	35	49				33	53	49			33	35	50			
34 35	35	49				بلان	52	49			34	35	50			
35	35	50				35	53	50			35	35	50			
36	35	50				36	52	50			36	35	50			
37 38	35	50				37	53	50			37 38	35	50 50			
30 39	36 35	50 50				38 39	53 53	50 50			30 39	35 34	50			
40	35	51				40	53	51			40	34	50			
41	35	50				41	52	50			41	35	50			
42	35	49				42	52	49			42	35	50			
43	35	50				43	52	50			43	35	50			
44	35	50				ža 14	52	50			հե	يئز:	50			
45	35	50				45	52	50			45	34	50			
46	36	51				46	53	51			46	34	50			
47	35	51				47	52	51			47	34. 34.	50			
48	35	50 56				48	53	50			48 49	34 34	51			
49 50	35 35	50 51				49 50	52 52	50 51			49 50	34 34	51 50			
<u></u> ,	37	74				0	16						,,,			
Remark		o probl	. <b>ems</b> vi	ere inc	urred.	Rema	rks :	No prot	lems were i	ncurred.	Remai	rks :	No prot	lens v	vere in	curred.

TABLE 2. Mechanical Operatics Data (Continued)

Gentry No. 3 and ' Load Weight: 15,500 lb Maximum Height Load Lifted: 62 in. Power Pack No. 2 .

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 Gentry Bo. 1 and 2
 Gentry Bo. 3 and h

 Load Weight: 4060 1b
 Lo.4 Weight: 16,000 1b

 Maximum beight Load Lifted: 62 in.
 Maximum Beight Load Lifted: 62 in.

 Power Pack Bo. 1
 Power Pack So. 2

 Ambient Temperature (\*P):
 Ambient Temperature (\*P):

 High, 93; Low, 81.7
 High, 93; Low, 51.7

 Bumidity (\$): High, h7; Low, 18
 Humidity (\$): High, 91; Low, 79

 Soil Temperature (\*P): High, 91; Low, 79

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- รบ	ne (sec)	Pressure	(hua) -	Time (dec		Pressure			Pressure
		Up Lown Idle	Cycle No.	11.00 Up	Down	Up	<u>Down</u>	1410	Cycle <u>Time (sec.</u> <u>Variation</u> <u>No.</u> Up Dove Up Dove Idle
52	49	Incorrect readings	1	36	49	1200	1100	1700	1 51 49 Incorrect reading
51	49	_	2	35	49				2 51 49
51	49		3	35	48				3 52 48
52	49		4	34	49				3 52 48 4 51 49
51	49		56	34	49				
52	49		6	34	49				3 51 49 6 51 49
52	49		7	34	49				7 52 69
52	49		8	34	49				8 51 49
52	49		9	35	49				9 52 hg
52	49		10	34	49				10 53 49
52	49		11	34	50				11 52 50
53	49		12	34	50				12 53 50
52	49		13	35	50				13 54 50
53	49		14	35	50				14 53 50
53	49		15	34	50	1000		1600	15 54 50
53	49		16	35	49				16 All readings
53	49		17	3 પં	50				17 approximately
53	50		18	34	50				18 the same.
54	50		19	34	50				19 A
52	50		20	34	50				20
52	50		21	34	50				21
53	49		22	34	50				22
53	50		23	34	50				23
53	49		24	34	50				24
53	49		25	34	50				25
50	49		26	34	50				26
52	49		27						27
53	49		28						28
53	50		29						29
53	49		30						30
53	49		31						
54	50		32						32
53	50		33						33
54	50		34.						34
54	50		35						35
54	50		36						35
54	50		37						30 37
54	50		38						31
بالر	50		39						30
54	50		40						40
54	50		41						40   141
54	50		42						41 42
54	50		43						
54	50		44						43 44
ŚĻ	50		45						
54	50		46			0900		1700	45 .6
54	50		40			0,00		1100	
54	51		48						47
54	51		40						48 All readings
54	50			21.	60				49 approximately
			20	يىر	20				ou the same.
urks: i ty pres	Pressure	gage worn and diving dings. Iber was -2).	50 Remar	34 ks: N	50 (v prob)	lezs i	ncurred	l <b>.</b>	50 the same. Remarks: Pressure gage wor giving faulty readings. It replaced.

Gentry No. 1 and 2 Load Weight: 3560 lb Harimon Bright Load Lifted: 62 in. Funner Peck No. 1 Subiest Temperature (\*?). Righ, 100; Low, 98 Sumifily (\$): Righ, 31; Low, 28

Pressure Qrela <u>fine (poc)</u> <u>Veriation</u> <u>No.</u> Up Doom Up Aven tile 28 28 1000 1100 1600 1700 prok je 30 wee incurvel.

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THREE 2. In vanical Operation Data (Continued)

Gestry Bo. 3 and b Load Weight: 16,500 lb Maximum Beight Load La. Yod: 62 in. Power Pack Pr. 2 Andiet: comparature (\*P): Aigh, 100; Low, 98 Bunidity (\$): High, 36; Low, 25

Contry Bc. 1 and 2 Load Weight: 3050 1b Hazimum Reight Load Liftof: 62 in. Pyper Pack Bc. 1 Ambient Temperature (\*P): High, 107; Low, 98 Humidity (%): High, 38; Low, 16

Cycle	- 21.me	(	Pressure Variation	Crele	91 ma	(		<b>100</b> 01219	
Bo.	. <u>5</u>	Dava	Un hous Idle	10.	Ů,	(pec) Down	<u>Ve</u>	<u>Dova</u>	Idle
1 2	54. 54.	51 50		1	33	50	1000	1100	1700
ŝ	- <del>7</del>	50		2	33 33	49 48			
3	55	50		2	33	50			
5	55	50		5	33	50			
	55	50		6	33	50			
7	55	50 50		7	33	50			
9	55 55	50		ხ 9	33 33	50 50			
10	ŝŝ	51		10	33	50			
ш	54	50		ü	<u>3</u>	18			
12	54	51		12	34	46			
13 14	54 55	51		13	3	49			
15	55	51 51		14 15	34. 34.	49 49			
16	55	51		16	34	49			
17	55	51		17	34	49			
13	55	51		18	34	50			
19	55	50		13	33	50			
20 21	55 55	30 50		20 21	33	50			
22	55	ŝõ		22	33 33	50 50			
23	55	50		23	33	50			
2t	55	50		24	33	50			
25 26	. 55	50	L .	25	33	50			
2.		nt na bì Blisto		25	34	50			
28	VICE I			27 28	3% 3%	50 50			
8		<b>∧</b> E		29	33	50			
30		;		30	32	50			
37				31					
22				32					
21 A 22 A				33 34					
35		•		35					
36				36					
37		÷		37					
× B				38					
- <b>P</b>				39 10					
41				41					
+2		1		12					
43				43					
44 45				24 1					
44				45 46					
•		4		47					
4.5				15					
3ý j		in as al		-9					
5	*11773	1 3000	200 L	50					
<b>June</b> :	nta: I	-	Oyule Io. 17 and 25.	<b>Inner</b>	bs: 1	io prob	leme -	are in	carred.
Sig State	tt atl	lask b	agen on Sentry Bo. 4.						
	728 22	ntag D	on apper cylinder at						
			spoint. Lask could						
			199 15-8). Pressure - ing family reading.						
		ىتودى د							
		-							

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TABLE 2. Mechanical Operation Data (Continued)

nindrati soro og

Gentry Ho. 3 and 4 Loud Veight: 17,000 1b Maximum Height Loed Lifted: 62 in. Power Pack Ho. 2 Ambient Tumpersture (\*7): High, 107; Low, 98 Humidity (\$): High, 38; Low, 18 Gantry No. 1 and 2Gantry 3 and 4Load Weight: 2560 lbLoed Weight: 17,500 lbMaximum Height Load Lifted: 62 in.Naximum Height Load Lifted: 62 in.Power Pack No. 1Power Pack No. 2Ambient Temperature (\*P):Mabient Temperature (\*P):High, 105; Low, 97High, 105; Low, 97Humidity (\$): High, 50; Low, 36Humidity (\$): High, 103; Low, 95Boil Temperature (\*P): High, 103; Low, 95

Cycle	**	(205)	Pressu Veriet		Cycle	74 ma	(aec)				Cycle	Time	(			_
No.	<u>Ve</u>	Denn	Up Down		No.	<u>V</u> a	Down	<u>Up</u>	Down	1410	<u>No.</u>	5	Dern	<u>Up</u>	Denta	Idle
1	54	50 48			1	32	49	1000	1100	1790	1	52	47	1700	1200	1700
2	5h				2	32	48				2	52	48 48			
3	54 54	48 50			3	32 32	49 49				3	52 52	48			
	3	50			-	32	50					53	48			
56	さいいいいいいいいいいいい	ŝõ			56	33	50				5	53	18			
7	54	50			Ť	32	50				7	55	48			
8	54	50			8	33	50				8	54 54	48			
9	54	50			9	33	50				9	54	49			
10	54	50			10	32	50				10	54	48			
11 12	54	48			11	32	49				11	53	49			
12	52	48			12	33	49 49				12 13	54 57	52 52			
13 14	23 51	49 49			13 14	33 33	50				13	55	51			
15	5	49			15	33	50				15	55			1100	
15 16	- G	49			16	33	50				16	55 54 56	50 51			
17	54	49			17	33	49				17	56	51			
18	Śł.	50			18	33	49				18	55	51			
19	54	50 50			19	32	50				19	55	51			
20	54	50			20	32	50				20	57	51			
21 22	55	50			21	32	49				21	58	51			
22	54	50 50 50			22	32	49				22	56	51			
23 24	55	50			23	33	50				23 24	56 55	51 52			
	55 81	50			24 25	33 33	50 50				24	22	52			
25 26	***	50 50			26	33	49				26	55 56	51 51			
27	- C	ŝ			27	33	49				27	54	51			
28	55	50 50 49			28	33	50				28	55	52			
29	55	50			29	32	50				29	54	51			
30	55	50 50			30	32	50				30	54	50 49	1500		
31	55	50			31	32	50				31	56				
32	55	50			32	33	49				32	55	51			
33	55	50			33	33	49				33	55	51			
34	55	49			34	32	50				34	54	50			
35	55	50			35 36	32 32	50 50				35	54 55	50 51			
30 31	55 51	50 50			30 37	32	49				31 32 33 34 35 36 37	55	31			
90 II 92 33 45 25 66 77 86 89 40	54 55 56 56	49			38	33	50				38	54	47			
39	55	49			39	33	49				38 39 40	53	50			
40	56	49			10	33	49					54 53	51	1600	1200	
41	56	49			41	32	50				41	53	48			
42	56	50			42	33	50				42	55	47			
42 43 44 45 46	57 58 58 56	50			43	33	50				43	- 54	46			
44	58	50			44	32	49				եե	54	47			
45	>0	50			45	33	49				45 46	54 55	48 47			
40 47	20 54	50 50			46 47	33 33	50 49				40	22 54	47			
46 48	56 56	50 50			48	33 32	50				45	55	49			
49	<b>~</b>	ŝ			49	33	50				49	56	48			
50	56 56	ŝõ			50	32	50				50	55	49			
-			ens vere is	curred.	•	-	-	lemu	vere i	scurred.	-		-	lems	were is	curred.

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TABLE 2. Mechanical Operation Data (Continued)

 Centry Ho. 1, 2, 3 and 4
 Gentry Ho. 1, 2, 3 and 4
 Gentry Ho. 1, 2, 3 and 4
 Gentry Ho. 1, 2, 3 and 4

 Loed Weight: 18,000 lb
 Loed Weight: 20,000 lb
 Loed Weight: 28,500 lb
 Loed Weight: 28,500 lb

 Maximum Height Load Lifted: 62 in.
 Heximum Height Load Lifted: 62 in.
 Heximum Height Load Lifted: 62 in.
 Heximum Height Load Lifted: 62 in.

 Power Pack Ho. 1 and 2
 Power Pack Ho. 1 and 2
 Power Pack Ho. 1 and 2
 Power Pack Ho. 1 and 2

 Amblent Temperature (\*7):
 Amblent Temperature (\*7):
 Amblent Temperature (\*7):
 High, 93; Low, 84

 Humidity (\$): High, 32; Low, 28
 Humidity (\$): High, 71; Low, 54
 Humidity (\$): High, 64; Low, 60

 Boil Temperature (\*7): High, 109; Low, 94
 Boil Temperature (\*7): Kigh, 97; Low, 85
 Boil Temperature (\*7): High, 88; Low, 80

			F		·e					sure						Häure	
Cycle	Time (			/eristi				e (sec)		istic				(gec)		ristio	
10.	<u>Up</u>	DOWN	<u>_</u>	Down	Idle	No.	Up	Down	Up	0048	THE .	No.	У <u>р</u>	Down	<u>Up</u>	Down	Idle
Readiz	gs from	a Powe	r Pac)	lo. 2	2 only	Readi	ngs	from Pow	er Pack	Ho.	2 only			vailable	from	Power	Pack
,	42	48	1200	1200	1700	1	43	48	1200	1200	1700	10. ž	only				
1 2	42	48	17.0	1200	1100	2	42	48	1200	15.00	100	1	45	48	1300	1200	1700
3	42	48				3	42	50				2	45	49			-,
ũ	42	48				i,	42	50				3	45	50			
5	42	48				5	42	50				4	45	50			
6	43	48				6	43	48				56	45	49			
7	42	47				7	43	48					44	49			
8	42	50				9	43	48				7 8	հեր	50			
9	42	50				9	43	50				9	ելել Հեյել	49			
10	42	50				10	43	49				10	44	50 50			
11 12	42 42	50 50				11 12	43 44	50 50				11	44	48			
13	43	50				13	43	50				12	44	49			
14	42	50				14	43	50				13	45	50			
15	42	50				15	43	50				14	44	49			
16	42	50				16	43	50				15	44	49			
17	42	50				17	43	50				16	45	50			
18	42	51				18	43	50				17	44	49			
19	42	50				19	44	50				18	44	49			
20	42	51				20	44	50				19	45	50			
21	43	51				21	41	49				20	44	50			
22	43	51				22	44	50				21	45	50 48			
23	42	50				23	43	50				22	46 44	40			
24	42	51				24	43 43	50				23 24	44	40			
25 26	42 42	51 51				25 26	43	51 51				25	45	49			
20	42	52				20	43	50				26	45	48			
28	42	52				28	43	51				27	45	48			
29	42	51				29	43	51				28	45	48			
30	43	52				30	43	51				29	45	48			
31	42	52				31	43	52				30	46	48			
32	42	52				32	43	51				31	45	48			
33	43	51				33	44	50				32	45	48			
34	43	52				34	44	51				33	45	48			
35	42	51				35	եր	51				34	47	49			
36	41	52				36	44	52				35	հե 45	50			
37	41	52				37	եր թեր	53 51				36	45	50 50			
38	42	52				38	43 44	54 53				37 38	46	51			
39 40	42 42	52 52				39 40	43	53				39	48	51			
40	43	53				41	43	52				40	46	51			
42	42	53				42	44	52				41	47	51			
43	43	52				43	44	52				42	47	51			
2,4	42	52				44	43	52				43	48	51			
45	43	53				45	44	52				ելել	47	52			
46	43	52				46	44	52				45	47	52			
47	42	52				47	43	52				46	47	51			
48	43	52				48	44	52				47	47	52			
49	42	52				49	43	52				48	47	52			
50	43	53				50	43	52				49	46	52			
						Demo		No	1			50	46	52			

Remarks: No problems were incurred.

Remarks: No problems were incurred.

Remarks: No problems were incurred.

Gantry No. 1, 2, 3 and 4 Load Weight: 25,000 lb Maximum Height Load Lifted: 62 in. Power Fack No. 1 and 2 Ambient Temperature (°F): High, 96; Low, 90 Humddity (%): High, 62; Low, 54

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TABLE 2. Mechanical Operation Data (Continued)

Gantry No. 1, 2, 3 and 4 Load Weight: 27,5.3 lb Maximum Height Load Lifted: 62 in. Power Pack No. 1 and 2 Ambient Tempersture (\*P): Nigh, 101; Low, 87 Humidity (\$): Nigh, 54; Low, 50 Gastry No. 1, 2, 3 and 4 Load Weight: 30,000 lb Maximum Height Load Lifted: 62 in. Power Pack Ho. 1 and 2 Ambient Temperature (\*F): High, 94; Low, 92 Humidity (\$): High, 66; Low, 62 Soil Temperature (\*F): High, 89; Low, 88

ycle Mo.	<u>Time</u> Up	(sec) Dova		Pressu Jariati Down	on	Cycle No.	<u>Time</u> Up	(sec) Dova		essure ristic Down	1414	Cycle No.		Down		essure ristio Down	
	<b>61</b> EVI	ilable					ngs a	vailable				Readi		veilable			
												<b>N</b> o. 2	•				
1 2	49 48	50 50	1400	1200	1700	1 2	48 49	50 50	1500	1200	1800	1	52	49	1600	1200	1800
3	46	49				3	49	50				2	52 52	49			
ъ́	47	50				1	49	50				1	53	49			
5	47	50					49	49				5	51	49			
6	47	50				5	50	50				6	52	50			
7	48	50				7	52	<b>5</b> 0				7	52	49			
8	47	50				8	52	50				8	52	50			
9	46	50				9	53	49				9	53	50			
10	45	51				10	53	50				10	53	50			
11 12	46 46	50 50				11 12	53 53	51 51				11 12	52	50			
13	47	51				13	53	50				12	53 53	50 50			
14	48	50				14	53	50				14	52	50			
15	47	51				15	54	50				15	53	50			
16	48	51				16	53	51				16	53	50			
17	49	51				17	53	50				17	54	50			
18	49	52				18	52	50				18	53	50			
19	49	51				19	52	50				19	54	50			
20	49	50				20	53		1400	1100	1700	20	54	50	1400		1700
21	47	48				21	53	51				21	54	50			
22	40 47	48				22	54	51				22	54	50			
23 24	48	49 50				23 24	54 54	51 52				23 24	54 54	50			
25	47	49				25	54	51				25	2** 53	50 50			
26	47	50				26	53	51				26	53	50			
27	47	50				27	53	51				27	53	50			
28	47	50				28	53	50				28	53	50			
29	48	50				29	53	51				29	53	50			
30	47	50				30	54	51				30	53	51			
31	48	50				31	54	51				31	53	51			
32	47	50				32	54	52				32	53	51			
33	47	51				33	54	52				33	53	51			
34 35	49 48	50 50				34 35	54 54	52 52				34u	53	52			
36	47	49				35 36	54 54	51				35 36	53 53	51 52			
37	49	50				37	54	51				37	53 54	52			
38	47	50				38	54	51				38	55	51			
39	47	50				39	Ś	52				39	54	51			
40	48	50	1300	1100		40	54	51				40	54	52			
41	48	50				41	54	51				41	54	52			
42	49	50				42	54	52				42	54	52			
+3	48	49				43	54	51				43	53	51			
la la	48	49				կել	54	51				հեր	53	51			
45	47	50				45	54	51				45	53	51			
46 47	47 47	50 50				46	54	51				46	53	51			
48 48	41	50				47 48	54 54	51 51				47 48	54 54	51			
49	47	51				40	24 54	51				40	54 54	51			
	46	50				49 50	24 54	51		1200		50	54 54	50 51			
50																	

.

TABLE 2. Mechanical Operation Data (Concluded)

Gentry Ho. 1, 2, 3 and 4 Load Weight: 32,500 lb Maximum Height Load Lifted: 62 in. Fower Pack Ho. 1 and 2

Gentry Ho. 1, 2, 3 and k Load Weight: 35,000 lb Maximum Height Load Lifted: 62 in. Power Pack Ho. 1 and 2 Ambient Temperature (°P): High, S6; Low, 9k Humidity (\$): High, 44; Low, 41 Soil Temperature (°P): High, 91; Low, 88 On the second

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Down           available           1y           3         50           3         50           3         50           3         50           3         50           3         50           2         51           2         51           3         50           2         51           3         50           2         51           3         50           3         50           3         50           3         50           2         50	Up from 1600	Down Power 1200	Idle Pack 1800	No. 1 2 3 4 5 6 7	2 only 54 54 54 54 53 53	<u>Down</u> vailabl 49 51 51 50 50 50 51	<u>Up</u> • from 1700	Down Power 1100	<u>Idl</u> Paci 1700
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TABLE	3.	Manual	Operational Data	
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	Max. Height		Tim	Require	ed (min/	ser)		
Lcad	Load	Cycl	le 1	Cycl	le 2	Cycle 3		
<u>Wt (1b)</u>	Lifted (in.)	Up	Down	Up	Down	Up	Down	
3,000	76	3/14	2/5	3/35	2/20	3/30	2/15	
6,000	76	4/40	3/15	4/25	3/10	4/50	3/20	
9,000	76	5/45	3/0	6/20	3/10	6/10	3/5	
12,000	76	6/0	3/35	6/14	3/20	6/25	3/38	
15,000	76	9/20	3/\$8	9/40	3/55	9/25	3/50	
17,500	76	10/15	4/30	10/20	4/45	10/18	4/20	

Item: Gantry Crane Power Pack	Specification: MIL-STD-461/462
Model: YAC32-1	Test Date: 15 August 1968
USA Reg Bo.: G000055	Test Area: 66 Percent Slope
Kamufacturer: Continental Notors Corp	Test Equipment: AN/URM 85

## Radiation Test - DE<sup>2</sup> Class III C

Freq				Freq			
Hes	A	<u>P</u>	<u> </u>	Hcs	<u> </u>	<u> </u>	<u>.</u>
0.15	71+	89	71+	110	25	58	47
0.35	71+	85	71+	120	27	57	49
1.5	71+	78	71+	130	28	57	54
3	71+	75	71+	140	31	56	58
3 5 8	57	72	59	150	34	56	52
	58	70	61	160	35	56	58
12	63	68	63	170	34	56	54
15	65	67	65	180	34	55	45
20	40	65	53	190	33	55	44
23	43	65	¥9	200	31	55	48
27	51	64	51	220	30	55	53
30	33	64	51	240	బ	56	71+
35	38	63	49	260	20	57	71+
38	35	<b>6</b> 3	51	280	23	58	71+
40	33	62	51	300	35	58	71+
45	30	62	52	350	2 <del>€</del>	60	71+
5û	28	61	59	400	23	61	71+
55	31	61	53				
60	36	61	44				
65	33	60	39				
70	46	60	45				
75	45	59	57				
80	39	59	46				
85	33	59	48				
90	32	58	52				
95	32	58	49				
100	31	58	48				

\*Decibels above one microvolt per megacycle of bandwidth A - Ambient noise level P - Passing limit a - Interference noise level at ambient

Remarks: As specified in MIL-STD-461; the limit for class IIIC items in the applicable frequency range of 0.15 to 400 MHz shall be relaxed by 20 db.

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APPE	NDIX II. FINDINGS	
Requirements	Source	Degree of Compliance
Individual components of the system shall be sufficiently lightweight to enable carry- ing for short distances and loading by four men onto a military vehicle.	Paragraph 2c, AMCTCM Approved TC	Did not meet requirement (Test No. 1 and Para. 1.1 App III).
No component or group of components of the system shall be of such a size as to prevent air transport by cargo aircraft in accordance with applicable portions of Appen- dices A and B of AR 705-35.	Approved TC	Met requirement (Test No. 1).
Component parts of the system must comply with the require- ments of approved specifica- tions (federal, military and/ or industry), and be made cor- rosion resistant through use of applicable standard methods an materials.	Approved TC	Did not meet requirement (Test No. 1 and Para. 2.1, App III).
The device must have a lifting capacity of 17,500 pounds and when used in pairs as a system must be capable of lifting a load measuring 108 inches high 110 inches wide, and 336 inches long, weighing 35,000 pounds, to a height which will provide a 60-inch ground clearance and will permit placement onto a ground transport vehicle up to 120 inches in width.	Approved TC	Met requirement (Test No. 2).
The system must be capable of manual assembly from shipping to operational condition, with out special tools or materials nandling equipment. Assembly time for a device (17,500-1b capacity) from removal from shipping skids to erection must be less than 1 hour, when using four men.	Approved TC	Met requirement (Test No. 2).
MOTHE TONT MEN.		29

Requirement	Source	Degree of Compliance
The system must have mechan- ical leveling provisions to insure stability in all direc- tions for all loads up to rated load on sloping terrain up to and including 5-degree slopes.	Paragraph 2c, AMCTCM Approved TC	Met requirement (Test No. 2).
The device, when assembled, must be capable of being man- propelled short distances over unsurfaced and non-trafficked areas in the vicinity of for- ward airfields.	Paragraph 2f, AMCTCM Approved TC	Requirement compliance could not be determined (Test No. 2).
The system must be capable of raising the rated load to a 60-inch height in approximate- ly 120 seconds, using self- contained gasoline engine operated power packages, to- gether with hydraulic control and lift components.	Paragraph 2g, AMCTCM Approved TC	Met requirement (Test No. 2).
The device shall be operable from a single control station.	Paragraph 2h, AMCTCM Approved TC	Met requirement (Test No. 2).
The system must be capable of manual operation if power is not available. With manual operation, the lift rate re- quirement of above is not mandatory.	Paragraph 2i, AMCTCM Approved TC	Met requirement (Test No. 2).
The system must meet the re- quirements of the current revision of Specification MIL-T-11748 (Signal Corps), "Interference Reduction for Electrical and Electronic Equipment."	Paragraph 2p, AMCTCM Approved TC	Met requirement (Test No. 2).
The system, when operated by its hydraulic power package, shall demonstrate with 95 percent reliability the capa- bility of performing a daily mission. A daily mission is defined as a total of 50	Paragraph 2b15, AMCTCM, Approved Operational Character istics and verbal request by Natick Lab	

Requirement

cycles (lifts of various load weights within the rated capacity. This implies 20 mission days as Mean Time Between Failures (MTBF). A failure is defined as that which prevents the unit from completing its assigned mission and cannot be repaired by the operator with the tools and materials provided within 30 minutes. Unscheduled organizational maintenance should not exceed 30 minutes during the performance of a daily mission. The total maintenance manhours will not exceed 10 percent of the operating hours on the basis of 8 hours of operation equal to 1 mission day. Total maintenance will include scheduled and unscheduled maintenance from operator level through direct support level.

The system must be capable of Paragraph 2j, AMCTCM Not tested. statically supporting twice the rated load without evidence of permanent deformation, when loaded at an attitude to 3 degrees in any direction from the vertical.

Approvea TC

Paragraph 2k, AMCTCM Not tested. The system must demonstrate sufficient reliability and Approved TC durability to lift 150 percent of its rated load to its fully raised height of 60 inches, for 50 cycles, with all overload safety devices rendered inoperable for duration of test. Paragraph 21, AMCTCM Met requirement (Test

Approved TC

The system must be easily maintained under field conditions. Components must be interchangeable between like items of the system. Maintenance costs must be a minimum for systems of this type.

31

No. 3).

Requirement

32

### Degree of Compliance

The system must be capable of Paragraph 2r, AMCTCM Not completely deter-operation and storage in tem- Approved TC mined (Test No. 3). peratures from -65°F to +125°F.

# APPENDIX III. DEFICIENCIES AND SHORTCOMINGS

1. Deficiencies

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Deficiency	Suggested Corrective Action	Remarks
1.1 One of the basic components of the gantry, the power pack, was too heavy to enable carry- ing for short distances and load- ing by four men onto a military vehicle.	None	If hydraulic oil were drained from power pack, weight would be within criteria.
2. Sh	ortcomings	
Shortcoming	Suggested Corrective Action	Remarks
2.1 The manual chain hoist chains corroded during testing.	Coat chains with non- corrosive material.	None
2.2 Baseplate wrench failed to operate properly.	Closer quality control be observed.	None
2.3 Oil seepage from hydraulic oil cylinder seals.	Better grade seals be used.	None
3. Sugges	ted Improvements	
Quality/Performance	Suggested Action	Remarks
3.1 Broken pressure gages.	Observe gages during service testing. If gages continue to fail, investigate possi- bility of replacement with a gage which will withstand high transient oil pressure.	None
3.2 Cable ends of the winch hoist cable are clamped and the loose ends taped. After 3 or 4 weeks the tape falls off and the wire cable is exposed which is a salety hazard.	Solder loose ends of cable.	None
3.3 Lifting loads with the manual chain hoist for 1 hour will cause blisters and open wounds on a man's hands.	Issue protective gloves to be worn when operating the manual chain hoist.	None

#### Quality/Performance

3.4 Stipping tape was wrapped around the vinch hoist crash baseles during shipment. During create a safety hazard it assembly of the test item, a small piece of tape was not re- . ands be tied (rather than soved between the vinch hoist busile wester and the winch heist, causing the safety brate to slip.

poorly coestructed.

### Secretic Action

**HERE'S** 

As small places of sage are - Kule easy to overlook and may is suggested that all loose taped) to the frame for ship-BES'-

	linnerko	011 leak (KPN Ly-2)	Lubrication and inspection of power pact.	(it teak (KPN LU-4).	Unacheduled Oil leek (KPR L5-1).	ULL Leak (KPN 45-4).	Hose off shelves (EPR 1.5-5).	Howe off shelves - Replaced 8 times - then disregarded ( mPH 19-5-2).	011 Leak (KPR L5-6).	Broken pressure gage (KPR L5-7).
202	"Ly pe Mai ntonance	Unechedul.ed	Boheduled	Unacheduled	Unscheduled	Umeneduled	Unscheduled	Unacheduled	Ипи¢ћефцаф	Unscheduled
TANUE L. MAINCONNES UNC	1.1 fm	a hr	25 hr	5 min	1.0 mLm	atm 4t	l, hr	2 hr to 3 hr 25 min	s hr 25 min	1.1 hr 7 min
T STINUT.	Active Maintenanco Time	L'o min	15 min	s min	J min	5 mtn	A min	at min	L hr 30 min (NOTK: No- placement time - 15 min)	∵ mtn
	Components and Helated (Juerationa	Manifold hone, quick fit disconnects (adjusted)	Org maintenance	Manifold nose quick fit disconnects (tightened)	Manifold hone quick fit disconnects (tightened)	Munifold home quick fit Alsconnects (repaired)	House assembly (adjusted)	Home wasambly (r.d.)แต่ted)	Hose assembly (repaired) NCTE: Repair created a harardous condition - item waa then replaced.	I'reнынга қақа ( <i>r</i> -µlanac)
	Guartery No.	2	2 4 7	4 2 1	1 4 1		1 <b>7</b> 5	4 <b>2</b> 5	न ब ह	-= 

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### APPENDIX IV. NAISTERANCE EVALUATION

Remarks	Broken pressure gage (EPR L5-7-2).	Broken pressure gage (EPR L5-7-2).	Unscheduled Oil leak (EPR 15-8).	Lubrication and inspection of power pack.
'l'ype Maintenance	Unscheduled	Unscheduled	Unscheduled	Bcheduled
Life Operation	10 hr 2 min	20 hr 32 min	20 hr	25 hr
Active Maintenance <u>11me</u>	5 min		U min	lý min
Components and Nelated Operations	3 k li P'resoure Kage (replaced)	3 & h Pressure gage (replaced)	Cylinder fitting gommev- tions (fightened, could not be repaired).	Org maintenance
Gantry No.	4 2 5.	न ब रा	2 7 7	17 <b>18</b> E

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TABLE 1. Maintenance Data (Concluded)

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### TABLE 2. Maintenance Times

Homenclature	Gantry <u>1 and 2</u>	Gantry 3 and 4	Total System
Operating hours	44.5	46.7	91.2
Active maintenance hours*	0.4	1.3	1.7
Maintenance ratio	0.009	0.028	0.019
Mean time between failures	44.5	46.7	\$2.2

"Includes unscheduled and scheduled inspection time.

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#### APPENDIX V. PHOTOGRAPHS

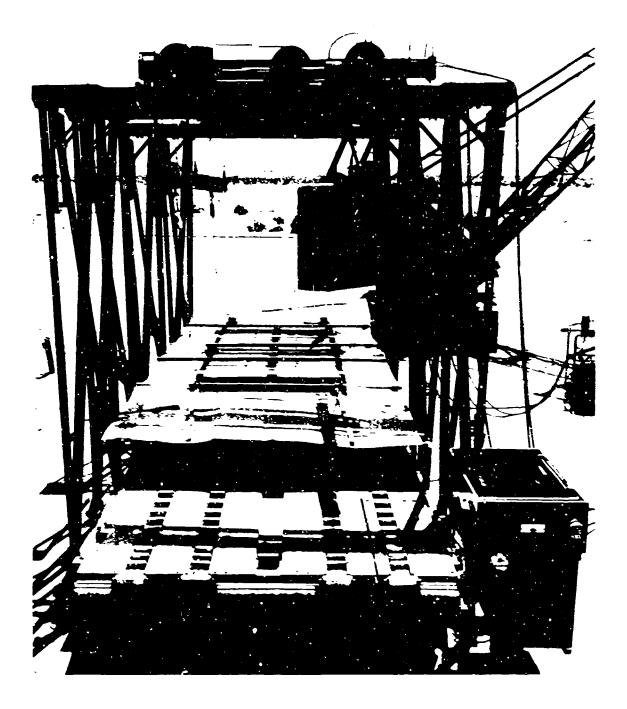


FIGURE 1. Aircraft loader (120 inches wide) driven under gantry system to pick up a 35,000-pound load which was extended to measure 336 inches long and 110 inches wide.



FIGURE 2. Man's hands after operating manual chain hoist for a period of 1 hour. Note blisters and open wounds.



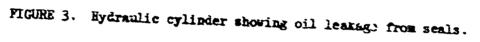




FIGURE 4. Broken gages which were replaced on the power pack during testing. At left, gage with the brass movement (note shavings). Gage wore thin. At right, gage with bronze movement. Gage broke.

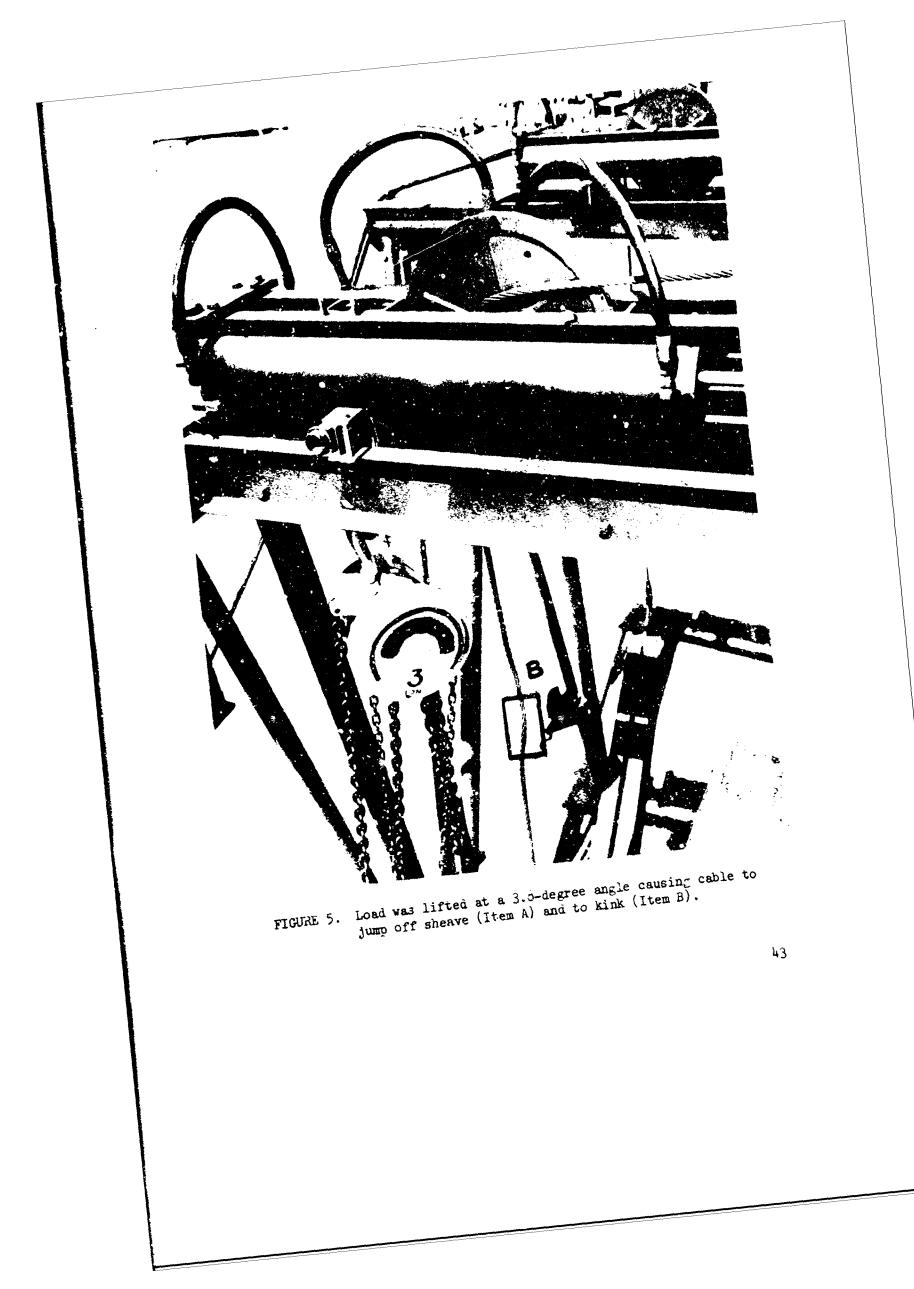




FIGURE 6. Cable ends of winch hoist cable.

1. Letter, AMSTE-BG, USATECOM, subject "Test Directive for Service Test of Gantry, Lightweight, Airdrop Rigging," 28 July 1966.

2. Letter, U.S. Army Natick Laboratories, subject "Preliminary Report of Engineering Design Test of Gantry, Lightweight, Airdrop Kigging," June 1966.

3. Letter, U.S. Army Natick Laboratories, subject "Revised Copy of Small Development Requirement and Technical Characteristics of Subject Gantry, as Amended by Minutes of In-Process Review Meeting, and as Approved by Indorsement from Hq, DA, OCRD," May 1968.

4. TM 5-2805-203-14, dated April 1965.

5. Letter, U.S. Army Natick Laboratories, subject "Gantry, Lightweight, Airdrop Rigging," 15 August 1968.

6. Maintenance and Operating Manual for the Portable Lightweight Lifting System for Preparation of Airdrop Cargoes, dated 7 March 1966.

7. U.S. Army Mobility Equipment Center, Draft Operator, Organizational, Direct and General Support Maintenance Manual (DTM 5-3950-205-14) for Gantry, Lightweight System for Preparation of Airdrop Cargoes, FSM (None Assigned), Natick Project NL-92.1, updated to March 1968.

8. USATECOM Regulation 385-2, Safety Responsibilities, 18 February 1963.

9. U.S. Army Airborne, Electronics and Special Warfare Board, subject "Service Test of Gantry, Lightweight, Airdrop Rigging, RDTE Project No. 1MI41812D18322A, USATECOM Project No. 4-5-7491-01 (AB 767)," June 1967.

10. USATECOM Regulation 385-7, Safety Confirmation, 18 December 1962.

11. USATECOM Regulation 700-1, Value Analysis in Materiel Testing, 18 February 1967.

(Security classification of title, body of obstract and index	NTROL DATA - R & D	verall report in classified)		
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UPPLEMENTARY NOTES	12. SPONSORING MILITARY ACTIVITY			
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The engineer test of the Gantry, Ligh uma Proving Ground from 20 May 1968 to The purpose of the test was to determ ervice testing. All testing was conducted under natur echnical characteristics of the test i est item reliability. The power pack ato a military vehicle (deficiency). roke, and the hydraulic cylinder leake It was concluded that the shortcoming pucluded that if only four men are ava rom the power pack, thus eliminating t acommended that the Gantry, Lightweigh esting.	Natick, Massachuset tweight, Airdrop Rigging, 30 August 1958. dime the suitability of th cal environmental condition tem were used as criteria was too heavy for four me The manual chain hoists of d oil (shortcomings). s were readily correctibil dilable the hydraulic oil the heavy weight deficience	ts 01762 was conducted by e test gantry for ns. The approved to determine n to carry and load orroded, the winch e. It was also could be drained y. It was		

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