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PERFORMANCE CHARACTERISTICS OF THREE PROPELLERS WITH VARYING PITCH DISTRI-BUTIONS ON AN INCLINED SHAFT

James G. Peck

Naval Ship Research and Development Center

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Naval Sea Systems Command

August 1974

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BFang	Bearing force angle, measured from the vertical BF_{ang} = arctan F_H/F_V [deg]
°0.7	Blade-section length at 0.7 radius [ft]
P	Propeller diameter [ft]
F _H	Horizontal force, measured perpendicular to the shaft [lb]
F _v	"Vertical" force, measured perpendicular to the shaft [lb]
F _{BF}	Bearing force $F_{BF} = \sqrt{F_H^2 + F_V^2}$ [1b]
J ·	Advance coefficient $J = V/nD$
K _{B⊦}	Bearing-force coefficient $K_{BF} = F_{BF} / \rho n^2 D^4$
κ _L	Lift coefficient $K_L = L/\rho n^2 D^4$
κ _Q	Torque coefficient $K_Q = Q/\rho n^2 D^5$
К _т	Thrust coefficient $K_T = T/\rho n^2 D^4$
L	Lift force $L = T' \sin \alpha + F_V \cos \alpha$ [lb]
n	Revolutions per second of unit time
p	Ambient static pressure [lb/ft ²]
p _v	Ambient vapor pressure {lb/ft ² }
Q	Torque [ft-lb]
т	Thrust in the horizontal plane $T = T' \cos \alpha - F_V \sin \alpha$ [lb]
T'	Thrust measured on the shaft [lb]
N	Speed of advance [ft/sec]
α	Angle of shaft inclination [deg]
η	Efficiency $\eta = TV/2\pi Qn$
ρ	Mass density of water [lb $-\sec^2/ft^4$]
σ	Cavitation number $\sigma = \frac{p_{-} - p_{v}}{1/2 \rho V^{2}}$

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ABSTRACT

A series of three commercial propellers with sysyematic pitch reductions at the hub were characterized at zero and 15 degrees shaft angle, over a range of cavitation numbers and advance coefficients. In addition to the usual thrust and torque forces, horizontal and vertical side forces were measured. These experiments showed that varying the pitch distribution of these propellers changed the type of cavitation on the propeller without significantly affecting the performance of that propeller.

ADMINISTRATIVE INFORMATION

This work was funded by the Naval Sea Systems Command, sponsored by PMS 300 under Funding Code 49501, Work Unit 1532-242.

INTRODUCTION

One of the most persistent _roblems which high-performance, small craft have encountered is propeller cavitation erosion in the blade root area. This problem was also observed on a series of constant pitch commercial propellers which were characterized on inclined shafts at the Naval Ship Research and Development Center (NSRDC).¹ The Naval Sea Systems Command (NAVSEA) requested that NSRDC investigate the effect of reducing propeller pitch at the hub on propeller blade root cavitation.

Two propellers were purchased with specific pitch reductions at 0.2R of 80 percent and 90 percent of the pitch at 0.7R. The design

¹Peck, J. G. and D. N. Moore, "Inclined-Shaft Propeller Performance Characteristics," NSRDC Report 4127 (Apr 1974)

pitch distribution of these propellers and of a constant pitch propeller from a series characterized earlier¹ are shown in Figure 1. The pitch of each blade of the three propellers was measured at several radial stations. An averaged value for the four blades of each propeller was calculated. These values are given in Table 1, and are presented as percent of pitch at 0.7 radius in Figure 1.

EXPERIMENTAL PROCEDUPE AND FACILITIES

Propeller open-water characteristics were obtained in the deepwater towing basin using a propeller boat with zero degrees shaft inclination. All three propellers were characterized in open water over a range of advance coefficients (J) from zero velocity to zero thrust loading. Reynolds number for the open-water tests ranged from 6.8 x 10^5 to 7.4 x 10^5 .

Cavitation characteristics of the propellers were obtained in the 36-inch variable pressure water tunnel using the right-angle shaft dynamometer. Cavitation experiments were conducted with a strut and shaft system upstream of the propellers simulating a typical full-scale environment. Tunnel water velocities for each propeller were established by setting thrust values at zero degrees shaft inclination equal to the thrust values obtained from the open-water characteristics tests at the same propeller advance coefficient. A water speed of 20 fps was used for all of the experiments. Reynolds number for the cavitation experiments ranged from 1.5 x 10^6 to 2.2 x 10^6 .

All three propellers were characterized at cavitation number values from 0.5 to 14.7 over a range of advance coefficients from zero thrust loading to maximum torque of the dynamometer. Characterizations were made at zero and 15 degrees shaft inclination. In addition to thrust and torque, forces perpendicular to the shaft in the vertical plane and in the horizontal plane were measured. Cavitation observations and sketches were made throughout the entire program.

PRESENTATION OF DATA AND DISCUSSION

The open-water characteristics data of the propellers were reduced to the usual nondimensional coefficients of thrust and torque. The characteristic curves of these propellers are presented in Figure 2. These curves show that, except when lightly loaded, the propellers with radially varying pitch distribution increasing toward the tip produce more thrust at the same advance coefficient but are less efficient than the constant pitch propellers.

During the cavitation characteristic experiments, forces perpendicular to the shaft and thrust and torque forces in the shaft were measured. In order to present the results in the usual coordinate system, the thrust and side forces were resolved into horizontal and vertical components as shown in Figure 3. The resultant thrust and torque data from the cavitation experiments were reduced to the usual nondimensional coefficients, K_T and K_Q , for each propeller. The lift coefficient K_L , bearing force coefficient $K_{\rm BF}$, and bearing force angle $BF_{\rm ang}$, were also computed from measured data. Efficiencies, K_T/J^2 and K_Q/J^3 were computed from the faired values of K_T and K_Q . All force coefficients are given in Tables 2 through 7.

The cavitation characteristics of the three propellers are shown in Figures 4 through 15. Figures 4 through 8 show the performance characteristics of each propeller for various cavitation numbers at zero shaft inclination and 15 degrees shaft inclination. Cavitation inception curves for the three propellers at 15 degrees shaft inclination are given in Figures 10 through 12. The cavitation inception curves represent the limiting values of sigma for cavitation at the section r/R under consideration; i.e., the area above the curves indicates there is no cavitation present at the given radius r/R. At a given

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advance coefficient and cavitation number, the type and location of propeller cavitation will be determined by the curves above this point on the chart. These curves clearly demonstrate the effect of pitch reduction at the hub on face cavitation. The pitch reduction causes face cavitation from the hub to 0.3 radius, at much higher cavitation numbers than on the constant pitch propeller.

Figures 13 through 15 show propeller efficiency versus propeller loading for each propeller at various cavitation numbers. These curves indicate the propeller efficiency which might be expected for a given propeller loading at a specific cavitation number.

Photographs of the constant pitch propeller and one of the reduced pitch propeller are shown in Figure 16. These photographs show that, for the same operating conditions, the constant pitch propeller has back cavitation over a larger extent of the blades, whereas the reduced pitch propeller has heavier face cavitation from 0.45 radius to the hub. Figure 17 presents sketches of the cavitation present on the three propellers at 15 degrees shart inclination for two propeller loadings. These sketches illustrate the decrease in back cavitation and increase in face cavitation on a propeller with pitch reduction at the hub compared to a constant pitch propelle. with the same pitch at 0.7 radius.

CONCLUSIONS AND RECOMMENDATIONS

These experiments show the changes in propeller cavitation which may be made by varying the radial pitch distribution of propellers having the same pitch at 0.7 radius. Reducing the pitch at the hub decreases back cavitation in that area but increases face cavitation.

The erosion damage caused by the different types of cavitation cannot be determined by the experiments. The performance of these propellers was not greatly affected by the pitch distribution. It is therefore recommended that constant pitch propellers continue to be used on highperformance small craft.

REFERENCES

1. Peck, J. G. and D. H. Moore, "Inclined-Shaft Propeller Performance Characteristics," NSRDC Report 4127 (Apr 1974)

	Tal	ole	1	
Pitch	Distribution	of	Propeller	Series

Propeller	4529	4615	4614
r/R	Averaged Measured Pitch Ins	Averaged Measured Pitch Ins	Averaged Measured Pitch Ins
0.2		7.637	7.360
0.3	10.117	8.300	9.120
0.5	10.104	9.588	9.565
0.7	10.175	10.217	10.144
0.9	10.289	10.776	10.505

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	INCLINATION	ANGLE =	v.non	FITCH RATIO =	1.000	\$15MA	= 14.700	
J	KTOUT	1060001	FFFIC	KT/J2	K0/J3	۴L	KRF	fir ANG
.6500	•2226	.375+	.6132	.5264	.1368	.0012	- 0024	60.5474
.7000	.199A	.3474	.6401	.4078	.1014	6013	0027	62. 1387
.7500	.1774	.3192	.6634	. 3154	.0757		0030	45.0422
.8000	.1551	.291.6	.6769	2424	-0570	.0013	.00.30	47 3445
- 8500	1326	. 2644	.6785	.1835	.0471	.0014	.00.17	5/46507 4 1 4700
	1000	2343	-070J	1340	0721	.0915	.0040	0.1.477
	•1043	+2340	*****	•] .)47	• • • • • • •	.0017	.0046	68.8037
+4260	.044H	•1443	• 7401	•0939	-0231	.0018	.0051	68.8050
1.0000	.0591	.1554	.6035	.0591	.0156	.0019	.0054	69.2896
1.0500	•0330	.1040	• 5055	•0299	•0094	.0019	.0059	71.0952
1.1000	.0079	.0649	•7125	•0065	•0049	.0019	.0071	74.8843
	INCLINATION	ANGLE =	0.000	PITCH RATIO =	1.000	SIGMA	= 3.000	
J	KTOUT	10×0001	EFFIC	KT/J2	KQ/J3	ĸĽ	KRF	HFANG
•6500	.2200	.3716	.6124	.5207	.1353	.0010	.0029	69.6555
./000	•2050	• 3457	•6511	•4123	.1008	.0007	.0030	75.7671
•7500	•1A1/	.3176	.6810	• 3722	.0753	-0005	.0032	80.7575
• #000	.15R1	•2884	•6967	.247L	.0544	.0006	.0035	79. A802
.R500	•1333	.2584	.6979	.1845	.0421	-0011	.0039	74.4412
.9000	.1076	.2241	.68/A	.1329	.0307	-0016	0444	44 0101E
.9500	.0816	.1845	.6641	.00/)5	.0215			67.7103
1.0000	.0557	1394	.4745	. 8567	+7412	•0014	.0050	77.1320
1.0500	-0296	.0921	-5372	•U737	•0139	-0016	.005/	/2.5410
1.1000	.0027	04.04		• • • • • • • • • • • • • • • • • • • •	-0080	.0014	.0065	79.0656
	•		• • • • • • • • • •	• 0022	•0037	.0019	-0070	74.1673
	INCLINATION	ANGLE =	0.000	PITCH RATIO	- 1.000	SIGMA	× 1.500	
J	KTOUT	1040001	EFFIC	KT/J2	KQ/J3	¥L.	KBF	HF ANG
-6500	.1562	.2864	.5640	• 3696	.1043	.0005	.0034	8].0644
-7000	1658	2992	.6173	.3384	.0872	.0006	.0040	81.7123
.7500	.1635	2975	.6560	, 2906	.0705	.0006	.0041	80.8255
	.1518	2814	-6857	.2372	.0551	.0008	.0041	77.4572
	.1331	.2546	.7072	-1843	.0415	-0012	.0042	74.0192
0000	1095	2188	7169	1 352	.0300	.0015	- 0046	70.5337
.4000	. 1077	1775	7044	.0918	-0207	0019	0050	LO 2044
.9700	+0020	1755	5470	.0551	.0136	0017	0054	40 6700
1.0000	+0751	•1355	.0470	. 0256	.0092	• • • • • • • • • • • • • • • • • • •	.0004	71 4441
1.0500	+02M2	.0974	*****	• • • • • •	+ U V O Z	.00/4	.00/4	71.4041
1.1000	.0045	• 000.3	.1.310	• • • • • • • • •		.0071	• VUNC	/1.0915
	INCLINATION	ANGLE =	0.000	PITCH RATIO	= 1.000	SIGMA	. .750	
L	KTOUT	LOKOOUT	EFFIC	KT/J2	KQ/J3	٣L	KBF	BFANC
.6500	.0754	.1670	-4672	.1785	.0608	0002	.0032	+00.4547
.7000	-0841	-1828	-5127	.1717	.0533	0002	.0034	-46.5191
.7500	.0917	1959	-5586	.1630	.0464	.0002	.0040	37,3538
	. 0963	.2071	.5917	.1504	-0405	.0009	.0046	87.7723
	AGE	,2127	. 6085	1122	0.346	.0016	.0051	92.7012
	00737 0970	2040	4043	1074	. 0201	.0021	.005	71.1846
• 4000	•U77V	1507	+0V02	74 7	. 4240		.006	53.0690
.9700	.0092	175	•2476	+0/0/	+ 4207	.0030	.007	58.7266
1.0000	.04/0	+1337	.3027	• • • • • • • •	.0135	00.V	0085	78.777
1.1000	•0101 ••0221	-0290	-1.3333	0183	.0022	.0052	.0094	53.8147
		ANGLE -	0.000	PITCH RATIO	= 1.000	SIGMA	= .500	
J	KTOUT	IOKOOUT	EFFIC	KT/J2	KQ/J3	K L	KBF	BFANG
							1544	A. 7408
.6500	.0397	.1004	.4021	.0927	•0367	.0004	.0070	17,2028
.7000	.0491	•1195	.4630	.1003	C46.0.	0001		J746VE0
.7500	.0554	.1300	.5089	.0985	80LV.	0007	.9945	-33.9411
.8000	.0604	.1450	•2298	•0943	.0283	0001	.0050	
.4500	.0617	.1571	.5311	.0854	•0256	.0007	.0058	1.7039
.9000	.0552	.1534	.5155	.0682	.0210	.0019	.0070	69.2888
.9500	.0375	.1233	.4597	.0416	.0144	.0035	.0085	103.5606
1.0000	.0086	.0675	.2040	.0086	.0057	.0050	.0097	76.3755
1.0500	0251	5900.	-6.7184	0228	.0005	.0059	.0094	18.6409
1.1000	0485	0113	7.5260	0401	0008	.0052	.0079	55.0316
				12<				

Table 2 - Performance Characteristics of Propsller 4529 at Zero Shaft Inclination

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		N ANGLE =	0.000	PITCH HATIL	. = 1.000	SIGMA	= 14.700	
J	KTOUT 1	INKGOUT	EFFIC	KT/J2	KQ/J3	R.	ĸŊF	BFANG
				74			0035 6	0. 7544
.6500	.7770	. 3001	1010	+53/4	.1340	-0012	.0038 7	3.0760
7500	.1794	- 3086	.6939	.3189	.0731	.0011	.0039 7	3.2014
. 8000	.1565	.2769	.7197	.2445	.0541	-0014	.0043 7	1.1506
.8500	.1330	.2423	.7426	-1841	.0395	.001A	.0044 6	9.1542
.9000	.1079	.2055	.7524	.132	.0282	.0020	.0057 6	9.2358
.9500	. 1809	-1669	.7330	.0896	.0195	.0770	.0066 7	2.1917
1.0000	.0523	.1252	.6648	.0523	.0125	.0017	.0074 7	6.5710
1.0500	.023A	.0763	.5207	.0216	.0066	*001H	.0081 /	(• 00000
1.1000	0016	.0111	2498	0013	.0009	•1F04	.0000 0	0
	INCLINATION	N ANGLE =	0.000	PITCH RATIO	= 1.000	SIGMA	s 3.000	
J	ктонт 1	0KQOUT	EFFIC	SL/14	KU/J3	¥1.	ĸĦŁ	BFA NG
.6500	.2279	.37yh	.0103	.5300	.1382	.0007	.0029	75.4496
.7000	.2039	.3514	•6455	.4162	.1026	.0008	.0079	74.1267
.7500	+1797	.3146	.6731	•3194	.0755	.0011	.0031	69.4587
.8090	.1546	.2837	.6936	.2415	• 0554	.0115	.00 75	65.4741
.8500	+1295	-74R4	.7050	.1792	.0405	.0014	.0045	64+C70C
.9000	•1037	4, 15.	./010	.1540	+ 4241	.0019	-0047	69.6126
.9500	.0762	.1//6	•0n/4 5736	• 97744	.0120	-0019	- 0066	73.1461
1.0000	•046M	.1277	+7/30	-0157	.0073	-0020	.007]	73.7539
1.0500	•9173 - 0073	.0410	- 1103	0060	.0031	.0025	.006*	68.2906
1.1000		10-10						
	INCLINATION	ANGLE =	0.900	PITCH RATIO	= 1.000	SIGMA 1	1.500	
J	KTOUT	1040001	EFFIC	KT/J2	K0/J3	KL.	KBF	PIP ANG
.6500	.1676	.30AH	.5615	.3968	.1125	.1005	.0033	A1.4845
.7000	.1719	.3181	.6022	.3509	.042/	.0005	.00.34	51.04VV
.7500	-1661	.3170	.6375	•2973	• 07.17	.0007	.00.33	71 3414
.8000	.1510	.2919	.0587	•2324	+ 1370	-0011	.0035	55.1661
- 4500	•15WW	· 26.38	• 77 ¥4 4 38 7	1960	.0314	.0022	.0048	52.1385
.9000	.1070		+0J07 6849	- 0009	.0219	-0024	-0056	63.4995
.9500	.0/30	1415	.4790	-8426	.0142	.0025	.0054	67.2883
1.0000	.04/0	.088)	1825	.0087	.0076	.0030	.0073	66.7631
1.0000	- 0302	.0255	-2.0761	0250	.0019	.0051	.0086	48.4228
1.1.000								
	INCLINATION	ANGLE =	0.000	PITCH PATIO	= 1.000	SIGMA :	• • • 750	
L	KTOUT	10×2001	EFFIC	KT/JZ	KU/J3	FL	601N	-96 1434
.4500	.0999	. 194H	.5301	.2703	.0/10	-,0007	•0014	-12.0405
.7000	.1925	.5050	.5634	• 2041	.0391		0032	40 4500
.7500	.1013	.2076	•2823	1234	.0410	.0016	.0041	99-6356
	.09A7	.2044	• 3424 • 4041	.1767	.0334	.0023	0050	77.5663
.4500	.0915	•2030 1874	• 5 079	.0064	.0257	-0012	.0057	49.4148
.9000	.0741	1614		-0611	.0177	.0043	.0067	34.996
.9500	• U221	.0958	.3654	.6220	.0096	.0054	.0077	44.3829
1.0000	- 0178	.0221	-1.3404	0161	.0019	•0066	.0046	51.4588
1.1000	0545	0593	1.6097	0450	0845	.0072	.0067	-28,4576
	THEL THAT I	ON ANGLE	= 0.000	PITCH RAT	10 = 1.00)0 \$1GM	, s ,501 	BFANG
J	KTOUT	10K00UT	EFFIC	KT/-12	KQ/J3	RL	- 682A	-24.5531
	.0598	.1420	.435P	.1416	.05)7	0000	.0025	-64.2589
.7865	.0676	.1550	• "	.1379	.0452		4629	-15.1510
7580	.9667	.1534	• >132	•1177	• • 365	-+************************************	.0039	47.5079
.4000	.0580	•14AH	.4960	.0905		_001R	.0055	87.4570
. 4500	.0441	.1384	.4310	• U hiv • Ajie		.0024	.0071	97.7087
.9000	.0255	•1173	•1112	26 44	.0006	.0020	.0042	84.8195
.9500	.0037	10622	• 4700		.0039	.0014	.0078	01.1014
1.0000	0204	•0341 •0341	-7,2693	0392	.0009	001A	.0051	31+1465
1.0780	AEAL	.0 194	-2.5935	0463 1	3 - 130	6935	041"	-1-01-1-01
1.1009	~ • • • • • •			· · · · ·				

Table 3 - Performance Characteristics of Propeller 4615 at Zero Shaft Inclination

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	INCLINATI	ON ANGLE =	0.000	PITCH RAT	10 = 1.00	U SIG	4A = 14.70	0
U	KTOUT	10K0001	EFFIC	KT/J2	KQ/J3	KL	KBF	BF ANG
.4580	.2234	.3668	-0316	.5300	. 1774	0017	0028	57.4109
.7000	.2006	.3405	.0560	.4093	-0907	0017	.0031	57.5566
.7500	.1781	.3132	.6790	.3167	.0742	-0017	.0035	60.7193
	.1549	.2794	.7048	.2421	.0547	-0018	.0039	62.7819
.8500	. 1305	.2423	.7286	.1806	.0394	.0019	.0043	63.9353
.9000	.104A	.204 0	.7361	.1294	.0280	.0020	.0046	64.5143
.9500	.0783	16H0	.7051	•0 8 68	.0196	.0025	.0052	64.8348
1.0000	•0511	•132P	•6117	.0511	.0133	.0025	.006u	65.0301
1.0500	.0274	.0897	-4179	•0203	.0077	.0129	.0055	64.8883
1.1000	0092	.016/	8661	0076	•0014	.0075	•007B	63.6885
	INCI. INATIO	N ANGLE =	0.000	PITCH RATI	0 = 1.000	SIGMA	= 3.000	
J	KTOUT	10KQNUT	EFFIC	K1/J2	K0/J3	۴L	KBF	BFANG
.6500	.2241	. 3792	.6113	.5304	.1381	.0008	.0025	70.3770
.7000	.2040	.3532	.6433	.4163	.1030	.000A	.0030	74.3377
.7500	1803	. 3224	.6074	. 3205	.0764	.0007	.0034	78.6387
	1555	.2897	.6832	.2429	.0566	.0007	.003H	79.6128
.4500	.1301	2565	.6861	.1800	.0418	.0009	.0040	77.5858
.9000	.1036	.2224	.6668	.1279	.0305	.0011	.0043	74.9620
.9500	.0755	.1870	.6103	.0R36	•0218	.0012	.0044	74.3102
1.0000	•0455	.1475	.4912	.0455	.0148	.0013	.0055	76.4501
1.0500	.0151	.1014	.2485	.0137	.00AB	.0014	.0063	78.5378
1.1000	0]24	.0453	4779	0102	+0034	•0017	•00/0	72.1520
1	NCLINATION	INGLE = 0.	.000 F	PITCH RATIO	= 1.000	SIGMA =	•750	
J	KTOUT	10K0011T	EFFIC	KT/J2	KG/J3	×٢	KAF	BFANG
. 6500	•0937	.1879	.5139	• 2209	.0684	0005	.0031	+13.7666
.7000	.0962	.1969	.5444	•1964	.0574	0003	.003A	-80.0924
•7500	• 0955	.2071	.5609	•1697	•0482	0001	.0045	-4.0161
.P000	•093A	.2067	•5777	•1465	.0404	.0002	.0051	59.4923
• • • • • • • •	.0790	.2022	.5956	•1232	.0329	.0005	.0055	90.8262
.9000	+U//U	1445	•002M	•0950	•0251	-0010	.0060	91.1750
1.0000	+ 1342	.1445	- 7000	.0000	•0199	.0019	.0065	75.9572
1.0500	0177	.0288	+1.0271	-0203	-0035	.0029	.007/	62.2532
1.1000	0453	0096	8.2429	0374	0007	• UU 5P	+00/0	58.03//
								24.0130
ĪN	CLINATION AN	KGLE = 0.0		TCH RATIO =	1.000	SIGMA #	1.000	DEANG
J	# TOUT	10K0001	EFF IC	K1/J2	NU/ J3	*L	n tir	
.6500	.1670	.2973	.5A10	• 3952	.10A3	0007	.00 75	-69,9498
.7000	.1734	.3107	•6217	.3539	.0906	0008		-80.0324
.7500	.1676	.3069	.6518	.29/9	.0/2/	0007	-005V	-95.881#
.R000	.1515	.2465	• 7/JI	•2107	• 0500		-0055	-19.2312
.8500	.1279	+2341	.0007	.1272	.0295	-0013	-0064	64-2581
.4000	.0997	.1764	-6045	.0772	.0203	-002A	-00H4	104.4582
.9700	.007/	1328	.4682	-0391	.0133	.0042	.0093	-20.9486
1.0500	.0076	0860	.1476	.0069	.0074	.0144	.0024	+01.2360
1.1000	0273	.0214	-2.2317	0226	.0016	.0017	0193	+03.5A37
	INCLINATION	ANGLE = (PITCH RATIO	= 1.000	SIGMA	= .500	
J	KTOUT	10KQOUT	EFFIC	KT/J2	K0/J3	ĸL	KHF	UFANG
.4500	.0601	.141)	.4403	.1422	.0514	.0005	.0030	81.0092
.7000	.0654	.1523	.4784	.1335	.0444	+0004	.00.56	NC+CUNY
.7500	.0667	.15%6	.4992	-1187	.0378	.0005	.00.12	71.J74J
.8000	.0638	.1617	.5023	• 0997	.0316	.0005	.0034	74,2654
.8500	.0563	.1557	•4887	.07/9	•VC34		- 00 JH	78.2373
.9000	.0427	-1373	•4456	• 0567	.0110	.0011	.0045	76.6286
.9500	•0195	.1014	.2910	.0205	.0642	.0015	.0053	74.0876
1.0000	0205	•V422	7,775		0640	.0020	.0002	70.0492
1.1000	UN4I - 2039		2,1161	JANA #		.0025	.0069	68.3903
1.000	*****	= 1001	C+1133	1004.1	4<			

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Table 4 - Performance Characteristics of Propeller 4616 at Zero Shaft Inclination

	INCL INAT	ION ANGLE =	15.000	PITCH RAT	10 = 1.000	SIGHA	= 14.78	9
J	KTOUT	10K00UT	EFFIC	XT/J2	K0/J3	¥L.	KBF	BFANG
	2121	3855	5870	. E 1 86	1606		4227	-13-6048
.7000	.1977	.3595	.6129	.4036	.1048	.0770	.0241	-12.7635
.7500	.1775	.3349	.6327	.3156	.0794	.0737	-0251	-11.5613
.8000	.1578	.3103	.6477	.2466	.0606	.0708	.0275	-10.3901
.A500	.1379	.2841	.6565	.1908	.0463	.0680	.0310	-9.4135
.9000	+1168	.2553	.6555	-1442	.0350	.0651	.0339	-8,6528
.9500	.0940	.2230	.6376	-1042	.0260	.0619	•0356	-8.0/20
1.0000	•0693	•1872	.5891	•0673	.0137	.0584	.0371	-7.5361
1.1000	-0160	.1464	.2581	.0132	.0081	.0507	.0654	-7.9951
	INCLINATI	ON ANGLE =	15.000	PITCH RAT	10 = 1.000	SIGMA	= 3.000	
J	KTOUT	10K00UT	EFFIC	KT/J2	K0/J3	KL	KBF	BFANG
. 6500	. 2107	3884	-5612	.4084	1414	A705	8335	-7 6800
.7000	.)970	. 3666	.5987	.4020	.1869	•U/97 0774	+4567	-7.00V8
.7500	.1786	-3408	.6254	.3175	.0808	.0740	.0257	-10.7821
.A000	·15A3	.3139	.6421	.2474	.0613	.0704	-0275	-10.7347
.4500	.1371	.2861	.6480	•1897	.0466	.0668	.029.	-10.0354
.9000	.1146	.2565	.6401	•1415	.0352	.0634	.0319	-9.0544
.9500	.0905	.2238	.6110	·1005	.0261	.0599	.03+8	-8.1532
1.0000	.0644	.1876	.5465	.0644	.0188	.0561	.0379	-7.6373
1.0500	.0377	.1492	.4226	.0342	.0129	.0522	.0410	-7.7088
1.1000	+0137	•1125	+2123	•0[13	•9985		• 94 30	-0172
	INCL INAT	ION ANGLE =	12+000	PITCH RAT	10 = 1.000	SIGMA	= 1.500)
J	KTOUT	10×00UT	EFFIC	KT/J2	KQ/J3	ĸL	KBF	BFANG
•6500	.1450	.2785	.5384	• 34 31	.1014	•0538	.0147	10.7725
.7000	.1525	.2886	.5889	•3113	.0841	.0608	.0193	5.2480
.7500	•1554	.2943	.6301	•2762	•0698	•0659	.0234	9895
. 4000	+1507	• 2872	.005]	•2355	• 0561	.06A1	.0269	-5.7514
.9000	-1189	+ 2000	.7193	• 1 9 1 2	.0434	- 0650	.0298	-8.3353
.9500	.0950	.2029	.7080	.1053	.0217	-0613	.0351	-9.0707
1.0000	.0690	.1687	.6510	.0690	-0169	.0574	.0381	-8.8743
1.0500	.0429	.1325	.5406	.0389	.0114	9537	.0414	-9.7574
1.1000	.0173	.0836	.3631	+0143	.0063	1499	.0449	-11.5552
	INCL INAT	ION ANGLE =	: 15.000	PITCH RAT	TIO = 1.000	SIGM	A = .75	0
J	KTOUT	10K90UT	EFFIC	KT/J2	KQ/J3	ĸL	KBF	BFANG
- 5500	.0759	.1888	.4158	.1796	.0657	.0256	.0054	17.6763
.7000	.0814	.2026	.4474	.1660		.0300	.0083	16.4611
.7500	.0866	.2144	.4820	-1539	• 9398	.0348	.0116	15,3385
.8000	•0900	•CCJI 2358	.5156	.1237	.0368	.0400	.0150	5.9127
.8500	.0873	.2193	.5376	-1016	.0301	.8486	.0257	-1.1174
.9500	-0675	.2013	.5069	.0748	.0235	.0501	.0312	-7.3837
1.0000	.0452	.1717	.4192	.0452	.0172	.0490	.0363	-11.4660
1.0500	.0185	.1339	.2304	.0168	.0116	.0457	.0406	-13.3180
1.1000	0064	.0962	1166	0053	.0072	.0420	.0437	-15.2050
	INCL INA	TION ANGLE	= 15.000	PITCH RA	TIO = 1.00	O SIGM	A = .5	••
J	KTOUT	10KOOUT	EFFIC	KT/J2	K0/J3	KL	KBF	BFANG
.6500	.0397	.1149	.3576	.0940	.0418	.0160	.0054	19.1620
.7000	.0443	.1255	.3928	.0903	.0366	.0172	.0053	14.8735
.7500	.0516	.1438	.4280	.0916	.0341	.0204	.0066	13.0623
. 8000	•056?	.1572	.4549	· 07/5		.0240	.0088	11.7246
. #500	•0557	.1014	• 4035	-0771	.0218	.0274		5,3984
.4000	.0444	.1207	_ 207A		.0174	.030/	.0226	1502
.7500	•VJY/ •N242	1304	2989	.0262	.0139	AAF0.	.0288	-6.0073
1.0500	_00QA	.1195	.1377	.0089	.0103	.0369	.0335	-9.7192
1.1000	0105	.0784	2351	0087	.0059	.0317	.0332	-7,4358
				1	5<	-		

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Table 5 - Performance Characteristics of Propeller 4529 at 15 Degrees Shaft Inclination

IADIE 6 -	Performance Shaft Inclin	Characteristics ation	of	Propeller	4615	at	15	Degrees
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	INCLINA	TION ANGLE	= 15.000	PITCH R	ATIO = 1.m	00 \$10	MA _ 14 .	
L	KTOUT	LOKGOUT	EFFIC	KT/J2	K9/.13	91 91C		700
45.A.A						~	RØP	BFANG
.7000	+2223	.3977	.5783	.5262	.1448	.0819	. 0229	-15 7000
.7500	1790	•3/1/	.5988	•4078	.1084	.0777	.0241	-14 5636
.8000	1503	• 3462	.0136	-3182	.0825	.0743	.0261	-13.0000
.8500	1243	+ 3221	.0257	.2473	• 9629	.0709	.0281	-11.5607
.9000	.1120	+6767	.0295	.1886	.0477	.0674	.0304	-10.4115
.9500	- 6953	+2026	.0120	.1383	•0360	.0636	.0329	-9.5514
1.0000	.0547	+23[3	•2212	.0945	.0270	.0593	.0356	-8.0354
1.0500	.0201	+177C	•4553	•0567	-0198	.0548	.0387	-0.7.15V
1.1000	. 0027	•1330	• 2019	.0255	.0134	.0504	.0419	-8.7734
			.4569	•0055	•0066	.0464	.0448	-10.029A
	INCL INA	TION ANGLE	= 15.000	PITCH RA	TIO = 1.0	10 616		
	~ * * • • • *		_				- J.V	
5	KIUUI	TOKDOUT	EFFIC	KT/J2	K0/J3	K1.	KBF	DE ANU.
.6500	.2163	3000	6337					Dr ANU
.7000	.2014	3717	•2/3/	+5119	•1420	.0A08	.0222	-4.8659
.7500	-1824	3451	.0039	•4111	.1084	.0791	.0245	-A. 7217
.4000	-1610	. 3156	•03VB	• 3242	.0818	•0758	.0265	-10.5810
.A500	-1380	2951	.0494	.2515	•0616	.0719	.0282	-10.7720
.9000	.1136	-2031	40348	-1910	• 0464	•0676	.0301	-9.9344
.9500	.0878	.2193	+0421	.1403	.0348	•0634	.0322	-8.8145
1.0000	- 0606	+6173	+0052	.0972	• 0256	.0590	.0347	-8 0743
1.0500	.0377	1007	•2338	.0606	-0181	.0547	.0375	-4 0912
1.1000	.0081	•13/0	•4062	.0302	•0118	.0505	-0407	-8.7612
	•••••	• • • • • • • • • • • • • • • • • • • •	+1241	.0067	•0067	.0469	.0437	-9.3008
	INCLINAT	TION ANGLE	= 15.000		710 - 1 -			
L	*TOUT	1.8400117		· LICH NA	110 = 1.000	IV SIGI	4A = 1.50	00
•		1040001	EFFIC	KT/J2	KØ/J3	KI.	KAF	BEANC
.6500	.1607	2083						OF ANG
.7000	.1657	3083	+22/4	.3804	.1086	-0604	. 0172	13 4453
.7500	.1643	3075	*2487	• 3391	•0898	.0655	. 0206	1364033
.6000	1556	- 30/3	.03/9	•2921	•0729	.0690	.024)	1.4228
.8500	1300	+C743 27A7	.0/26	•2431	• 0575	.0701	.0275	-3 46 22
.9000	.1187	+6/0/	•0AA3	.1937	.0441	.0688	.0305	-7.6633
.9500	.0937	+C37E	•/111	.1466	.0358	.0656	.0331	-/.04/3
1.0000	.0663	+EVED	•0499	-1038	•0237	-0614	- 0351	-10.0113
1.0500	0360	1143	-0462	.0663	+0163	.0569	.0386	-10.6117
1.1000	.0045	-1103	+230Z	•0335	-0100	.0522	.0419	-12 1162
		60J31	•1468	•0038	.0042	.0470	.0462	-15.8200
	INCLIN	ATION ANGLE	= 15.000	PITCH P	ATTO - 1.4	-		-13.02.00
,						510		750
5	KIUUI	TOKGOUT	EFFIC	KT/J2	KQ/J3	K 1.	KBE	
.6500	.0808	1709	4000				-114	BP ANG
-7000		+1/00	.4893	•1915	• 0622	.0278	.0063	10 4313
.7500	- 694.0	+1002	.5312	.1831	.0549	.0341	.0100	17.4311
.8000	. 6933	.1703	.5049	•1670	•0471	.0389	.0134	120/222
.8500	.0873	1020	,5929	-1458	•0391	.0422	-0166	2 1074
-9000	.0753	+1700	.6121	.1208	.0314	.0440	.0199	P1014
.9500	.0575	1400	.0144	.0929	•0241	.0446	.0237	-611799
1.0000	.0349	•1400	.5839	.0637	•0174	.0442	. 0284	-0.2014
1.0500	-0102	0734	490/4	.0349	.0114	.0427	.0115	-11+3102
1.1000	0122	0707	+ 2320	+0092	•0063	.0398	-0379	-10*0411
		**3*2		0101	+0023	.0344	.0390	-19.5020
	INCLINAT	ION ANGLE	= 15.000				•	1010004
			- 150000	FILL RA	10 = 1.00	U SIGM	A = .50	0
2	KTOUT	10KQOUT	EFFIC	KT/J2	K0/J3	KL.	KBF	RFANG
-6500	-0514	1345	305-					DI ANU
.7000	.0504	1883	• 3423	•1217	.0490	.0188	.0058	18.3419
.7566	-0677	1736	**173	+1192	• 0452	.0227	.0068	10.8549
.8000	- 4655	+1/63	.+3/8	•1125	.0409	.0277	.0098	4.7211
	.0622	+10CL	000	•1028	•0356	.0314	.0132	.9603
.9000	.0447	+1011 .1714	9C 0+.	1080.	.0296	.0328	.0163	8112
-9500	.0244	+1/14	*****	• 0001	• 0235	.0320	.0190	-1.6174
1.0000		.1267	• C733	•UZ12	+9178	.0295	.0215	-2.3599
1.0504	0224	.1431			-0129	.0258	.0233	-3.0555
1.1000	6614	- 0601	0219		• 9989	•0207	.0232	-2.0725
		• AGAT	-14216	-•0015	• 9960	.0129	-0189	4.6314

	INCLINATION ANGLE = 15.000		PITCH RATIO = 1.000		SIGMA = 14.700			
L	KTOUT	10KQOUT	EFF 1C	KT/J2	K@/J3	KL.	KBF	BFANG
-6500	.2189	. 3884	.5830	•2181	.1414	.0810	.02 2	-13.5094
.7000	.1971	. 3639	.6035	.4023	.1061	.0772	.0241	-12.3259
.7500	.1774	. 3429	.6176	•3154	.0813	.0740	.0260	-10.7703
.8000	.1572	.3191	.6272	.2456	.0623	.0708	.0260	-9.3467
.8500	.1353	.2907	.6294	.1872	.0473	.0672	.0302	-8.3020
.9000	•1113	.2583	.6173	•1374	•0354	.0633	.0326	-7.6955
•9500	.0856	.2234	•5797	.0949	.0261	.0592	.0353	-7.4679
1.0000	•0588	.1867	.5009	+0588	•0187	.0550	.0383	-7.5112
1.0500	•0309	.1465	• 3531	.0281	•0127	.0509	.0416	-7.7373
1.1000	•0019	.0965	.0341	•0016	•0073	•0469	•0453	-8.1480
INCLINATION ANGLE = 15.000		PITCH RATIO	PITCH RATIO = 1.000		SIGMA = 3.000			
J	KTOUT	10KQOUT	EFFIC	KT/J2	K0/J3	KL.	KBF	BFANG
.6500	•2159	.3776	•5916	•5111	.1375	.0810	.0225	-6.6045
•7000	.1995	• 3542	.6275	•4071	.1033	.0788	.0248	-9.6279
•7500	.1803	.3264	.6594	• 3206	•0774	.0755	.0267	-10.5978
. M000	+1596	.2977	.6827	•2494	.0581	.0718	.0285	-10.2386
.8500	•1375	.2690	•6918	•1904	•0438	.0679	.0304	-9.2993
.9000	•1138	•2394	.6807	.1404	•0328	•0639	.0327	-8.4377
• 9500	.0879	.2000	.6430	.0974	•0241	.059A	.0354	-8.1061
1.0000	+V398	•10/0	• 2078	• 0578	.0167	.0555	•0385	-8.4360
1.1000	+0302	+1100	• • 3 • 9	.02/4	.0100	+0508	•0418	-9.1231
1.1000		• (403	.0311	.0007	•0036	• 1420		-7.3124
INCLINATION ANGLE = 15.000		PITCH RATIO	\$IGMA = 1.500					
J	KTOUT	10KQOUT	EFFIC	KT/J2	KQ/J3	KL	KBF	BFANG
.6500	.1596	.3110	.5308	.3777	.1132	.0597	.0167	13.4178
.7000	. 1657	.3230	•5716	•3381	.0942	•0660	.0210	5.3333
.7500	•1654	.3253	.6070	•2941	.0771	.0702	.0251	-2.1391
.R000	•1562	.3153	.6308	•2440	.0616	.0710	.0284	-7.0964
.8500	. 1385	.2930	•6396	.1918	.0477	.0689	.0310	-9.4678
.9000	.1149	.2610	.6307	+1419	.0358	.0649	.0335	-10.2548
.9500	.0883	.222/	•5995	.0979	.0260	.0607	.0365	~10.//13
1.0000	.0606	.1817	+5312	.0606	-0182	.0570	.0402	-11.8831
1.0500	.0316	.1405	• 3764	.0207	.0121	.0530	.0441	-13.2488
1.1000	002/	•0997	0472	0022		•U450	.750	-12+3371
	INCLINATION	ANGLE = 1	5.000	PITCH RATIO #	K0/ 13	310MM -	KBF	BFANG
J	K1001	100000	Erric	7043	0734		.0074	16.3909
•6500	.0879	.2013	61CPe	.1017	.0674	.0361	.0109	13.6581
./000	.0939	+ 21/7	+4014 5133	.1751	.0544	.0411	.0144	9.7134
• /500	+U787 A900	2280	-5166	.1560	.0465	.0454	.0181	4.1701
	.0958	- 240)	.539A	-1326	.0391	.0486	.0222	-2.1941
.9000	.0846	.2326	.5210	-1045	.0319	.0502	.0269	-8.1039
.9500	- 0656	2127	.4660	.0726	.0248	.0499	.0319	-12.4520
1.0000	.0396	.1796	.3510	.0396	.0180	.0475	.0369	-14.9034
1.0500	.0100	.1353	.1231	.0090	.0117	.0434	.0412	-10.0/10
1.1000	0172	.0866	3469	0142	.0065	.0387	.0439	-21+3040
	INCLINATI	ON ANGLE =	15.000	PITCH RATIO	= 1.000	SIGMA	= .50) DEANG
J	KTOUT	10KOOUT	EFFIC	KT/J2	K0/J3	KL.	RUP	10 2236
.6500	.0638	.1750	.3773	-1511	•0637	1220.	• UUDI	10.22JU
.7000	.0742	.1979	.4177	+1514	•0577	•0284	.0007	11.0023
.7500	.0803	.2155	•4450	.1428	.0511	.0337	.UII7	5.8146
	.0811	.2256	•4578	.1267	.0441	.0373	*A19E	3.0170
.8500	.0757	• 5569	.4512	+1047	•0369	•0344	•V102	
.9000	.0637	.2189	.4170	•0787	.0300	.0403	A277	-10.5140
.9500	.0455	.2016	•3410	•0504	• 4235	•U=U=	.0232	-13.0932
1.0000	.0219	•1753	.1985	+0219	•VI/5	00377 0348	.0375	-14.5972
1.0500	0055	.1401	0651	-+0020	+VIZI	.0280	.0374	-17.7461
1.1000	0340	.0955	-•0229	17<	••••	4 V E UV	17917	

Table 7 - Performance Characteristics of Propeller 4616 at 15 Degrees Shaft Inclination









Figure 1a - Side View



Figure 1b - End View

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